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DIGITAL ANTENNA GONIOMETER AND METHOD

TO ALL WHOM IT MAY CONCERN:

BE IT KNOWN THAT DAVID A. TONN, employee of the United States Government, citizen of the United States of America, and resident of Charlestown, County of Washington, State of Rhode Island, has invented certain new and useful improvements entitled as set forth above of which the following is a specification.

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2 DIGITAL ANTENNA GONIOMETER AND METHOD 3 4 STATEMENT OF THE GOVERNMENT INTEREST 5 The invention described herein may be manufactured and used 6 by or for the Government of the United States of America for 7 8 Governmental purposes without the payment of any royalties thereon or therefore. 9 10 CROSS-REFERENCE TO OTHER PATENT APPLICATIONS 11 Not applicable. 12 13 14 BACKGROUND OF THE INVENTION Field of the Invention 15 (1)The present invention relates to antennas and, more 16 particularly, to a digital goniometer especially suitable for 17 controlling a steerable antenna pattern from two orthogonal 18 antennas over the VLF/LF communications band. 19 20 (2) Description of the Prior Art Submarine antenna systems may frequently utilize the VLF/LF 21 22 communications band (8kHz - 200 kHz) by means of two magnetic 23 loop antennas which may be referred to as a multifunction

The two magnetic loop antennas are preferably 1 antenna. identical and aligned perpendicular to each other in one 2 One loop is referred to as the Fore and Aft (F/A)3 housing. The other loop is referred to as the Athwartships 4 antenna. (ATH) antenna. The multifunction antenna does not rotate, and 5 there are null spots where the antenna gain is near zero. 6 Orienting this null spot toward a source of interference is 7 desirable to reduce the received interference. Likewise, 8 orienting the highest gain region of the antenna toward the 9 10 desired signal is also desirable. Currently, an analog 11 mechanical device is used to combine the F/A signal with the ATH signal in order to obtain the desired orientation. 12 The 13 mechanical device contains a stator for each loop and a rotor 14 that can be rotated to adjust the proportional contribution of each loop. The device employs Faraday's Law of Induction for 15 16 its operation. The mechanical device presently utilized is an 17 electromechanical resolver often known as a goniometer. While goniometers may be utilized to achieve other effects and 18 19 functions, a goniometer may also be utilized for steering a 20 figure-eight antenna pattern with two identical orthogonal antennas by mechanical means. 21

1 It would be desirable to provide a more accurate, 2 lightweight, compact, antenna goniometer that does not require 3 elaborate external drive circuitry and has no moving parts. 4 Patents that show attempts to solve the above and other 5 related problems are as follows:

U.S. Patent No. 5,402,132, issued March 28, 1995, to Hall 6 7 et al., discloses a direction finding system utilizing a single monopole/crossed slot antenna in combination with associated 8 9 electronic circuitry. Each of the four ports of the antenna are 10 connected to amplitude varying elements whose outputs are 11 combined by a power combiner. A microcontroller stores the signal. The microcontroller also generates the signals which 12 control the amplitude variation supplied by each of the 13 14 amplitude varying elements. Two configurations of the amplitude varying elements are provided by predetermined settings within 15 16 the microcontroller. The two settings are selected to configure 17 the antenna on two distinct reception patterns, each with a predetermined angular offset from a reference direction. In 18 19 operation the antenna will be configured for a first reception 20 pattern. The signals received will be detected, quantified, and 21 stored by the microcontroller which subsequently reconfigures 22 the antenna for a second reception pattern. A second signal is 23 received by the antenna in the second configuration and again is

detected, quantified, and stored by the microcontroller. The microcontroller determines the difference between the two signals and, based upon a predetermined calibration curve relating differences in the stored signals to the angle of arrival of the signals from the reference direction, determines the angle from which the signal arrived.

7 U.S. Patent No. 6,160,519, issued December 12, 2000, to Christian O. Hemmi, discloses a two-dimensionally steered 8 antenna system including a planar lensing system operable to 9 10 focus signals received from a plurality of ground-based cells. A 11 first steering system is operable to steer a beam for each 12 ground-based cell in a first direction by weighing signals 13 associated with the ground-based cell based on a position of the 14 antenna system relative to the ground-based cell in the first 15 direction. A second steering system is operable to steer the 16 beam for each ground-based cell in a second direction by 17 weighing signals associated with the ground-based cell based on a position of the antenna system relative to the ground-based 18 19 cell in the second direction.

U.S. Patent No. 4,150,382, issued April 17, 1979, to Ray J.
King, discloses an invention which provides a guided wave
antenna having a radiation pattern which can be controlled

electronically, by control signals derived from a computer or 1 any other suitable source. In this way, the directional 2 characteristics of the antenna can be adjusted and/or scanned 3 rapidly, without any mechanical manipulation of the antenna. In 4 one embodiment, a quided radio wave is launched along an antenna 5 surface having an array of elements which provide variable non-6 uniform surface impedance adapted to be controlled by electronic 7 signals. For example, each variable impedance element may 8 9 comprise a wave guide section having one end leading from the antenna surface. Each wave quide section may include a solid-10 state electronic reflection amplifier having characteristics 11 12 which can be varied by supplying control signals to the amplifier, to vary the magnitude and phase angle of the wave 13 reflected from the reflection amplifier. By changing the control 14 15 signals supplied to any particular reflection amplifier, it is possible to cause attenuation or amplification and phase shift 16 17 of the guided wave as it passes across the particular wave guide section. A wide variety of solid-state electronic control 18 elements may be provided along one or more surfaces of the 19 antenna. In another embodiment, a wave traveling in a closed 20 subsurface wave guide is coupled into the guided wave open 21 22 surface structure, using an array of wave guide elements

containing electronically controllable amplifiers and phase
 shifters.

U.S. Patent No. 5,714,961, issued February 3, 1998, to Kot 3 et al., discloses a directional planar antenna. The antenna has 4 an array of coaxial ring-slot radiating elements formed through 5 a conductive layer on a dielectric substrate. A number of 6 probes, coupled to the ring-slot elements, selectively excite a 7 separate resonant mode on each ring-slot element. The resonant 8 mode supported by a ring-slot element depends upon the geometry 9 of that ring-slot element. The resonant modes combine in the far 10 field to form a radiation pattern directional in azimuth and 11 elevation. By adjustment of the relative phase difference or 12 relative amplitude between the excited modes, the radiation 13 pattern can be steered. 14

The above-cited prior art does not show a suitable lightweight, compact antenna goniometer with non-moving components utilizing two orthogonal antennas that is operable for steering a figure-eight antenna pattern. Those skilled in the art will appreciate the present invention that addresses the above and other problems.

SUMMARY OF THE INVENTION

Accordingly, it is an object of the present invention to
provide an improved antenna goniometer.

1

It is another object of the present invention to provide an
antenna goniometer operable with two orthogonal antennas for
steering a figure-eight antenna pattern.

7 An advantage of a system in accord with the present 8 invention is an antenna goniometer that can be operated in a 9 fully automated manner that can be controlled from software.

In accordance with the present invention, a VLF/LF 10 communications antenna goniometer for a first antenna loop and a 11 second antenna loop is provided that is operable for controlling 12 an angular orientation of a figure-eight antenna reception 13 pattern for the first antenna loop and the second antenna loop. 14 The preferred embodiment shows first and second antennas which 15 are orthogonal with respect to each other; however, the 16 17 teachings of this invention can be applied by one of ordinary skill in the art to other antenna configurations. 18 The goniometer comprises one or more elements such as, for instance, 19 20 a first variable gain amplifier for a first signal produced utilizing the first antenna wherein the first variable gain 21 amplifier has a first output, a first data register for storing 22 23 a digital value related to a first gain setting for the first

variable gain amplifier such that the first gain setting is 1 related to the angular orientation, a second variable gain 2 amplifier for the second antenna for a second signal produced 3 utilizing the second antenna wherein the second variable gain 4 amplifier has a second output, a second data register for 5 storing a digital value related to a second gain setting for the 6 second variable gain amplifier such that the second gain setting 7 is related to the angular orientation, and a summing amplifier 8 9 for combining the first output and the second output.

Other elements of the goniometer may comprise a first digital data link for supplying the first gain setting to the first data register and a second digital data link for supplying the second gain setting to the second data register. In the preferred embodiment, the first gain setting is a sine function related to the angular orientation, and the second gain setting is a cosine function related to the angular orientation.

The goniometer may further comprise an inverter for
inverting at least one of the first signal or the second signal.
The summing amplifier adds the first output and the second
output together.

In operation, a method is provided for controlling an antenna reception pattern for the first antenna and second antenna wherein the method comprises one or more steps such as,

for instance, producing a first antenna signal utilizing the 1 first antenna, producing a second antenna signal utilizing the 2 second antenna, determining a first gain signal related to a 3 desired antenna reception pattern, determining a second gain 4 setting related to the desired antenna reception pattern, 5 6 digitally adjusting the first gain setting for the first antenna signal to produce a first antenna processed signal, digitally 7 adjusting the second gain setting for the second antenna signal 8 9 to produce a second antenna processed signal, and combining the first antenna processed signal and the second antenna processed 10 signal. Additional steps may include determining the first gain 11 signal and the second gain signal to produce a figure-eight 12 reception pattern having a desired angular orientation. 13

Other method steps may include providing that the first antenna signal and the second antenna signal are substantially in phase with each other to produce a figure-eight pattern. Additional steps may include utilizing software to adjust the first gain setting and the second gain setting.

In more detail, the method may comprise controlling a first magnitude of the first antenna signal with the first gain setting to produce the first antenna processed signal, and controlling a second magnitude of the second antenna signal with the second gain setting to produce the second antenna processed

signal and/or providing that the first gain setting is related to a sine function of an angular orientation of a figure-eight reception pattern, and providing the second gain setting is related to a cosine function of the angular orientation of the figure-eight reception.

6 In other words, a method is provided to produce a figureeight antenna pattern with a selected angular orientation 7 comprising one or more steps such as, for instance, providing 8 9 that a first antenna signal and a second antenna signal are substantially in phase with respect to each other, and digitally 10 controlling a first magnitude of the first antenna signal and a 11 12 second magnitude of the second antenna signal such that the 13 first magnitude is related to a sine function of the angular 14 orientation and the second magnitude is related to cosine 15 function of the angular orientation. Other method steps then 16 include combining the first antenna signal with the second antenna signal. The combining may further comprise summing the 17 18 first antenna signal to the second antenna signal. Preferably 19 the method comprises utilizing software to compute a first gain 20 setting for the first magnitude and a second gain setting for 21 the second magnitude, storing the first gain setting in a first 22 register and storing the second gain setting in a second

register. Preferably the method permits for dynamically
 changing the angular orientation utilizing software.

These and other objects, features, and advantages of the 3 present invention will become apparent from the drawings, the 4 descriptions given herein, and the appended claims. It will be 5 understood that above listed objects and advantages of the 6 invention are intended only as an aid in understanding aspects 7 of the invention, are not intended to limit the invention in any 8 way, and do not form a comprehensive list of objects, features, 9 and advantages. 10

11

12

BRIEF DESCRIPTION OF THE DRAWING

13 A more complete understanding of the invention and many of 14 the attendant advantages thereto will be readily appreciated as 15 the same becomes better understood by reference to the following 16 detailed description when considered in conjunction with the 17 accompanying drawing wherein corresponding reference characters 18 indicate corresponding parts and wherein:

19 FIG. 1 is a schematic diagram showing a pair of antennas 20 perpendicular to each other in a coordinate system; 21 FIG. 2 is a schematic diagram of a digital antenna 22 goniometer in accord with the present invention; and

FIG. 3 is a block diagram of a goniometer operable for
 steering a VLF/LF figure-eight antenna pattern.

3

BRIEF DESCRIPTION OF THE PREFERRED EMBODIMENTS Referring now to the drawings, and more particularly to FIG. 1, there is shown a circuit diagram of antenna goniometer 10 in accord with the present invention. Antenna goniometer 10 provides a more lightweight, compact antenna goniometer which does not require elaborate external drive circuitry.

10 Furthermore, antenna goniometer 10 has no moving parts and is 11 fully controllable by software from a bus such as a TTL parallel 12 bus.

FIG. 2 discloses antenna system 12 with electrically short 13 electric loop antennas 14 and 16 perpendicular to each other and 14 mapped onto a coordinate system which utilizes spherical 15 coordinates as indicated, such that antenna 16 is aligned with 16 the z-axis and antenna 14 is aligned with the y-axis. This 17 arrangement represents two crossed electric loops, but because 18 19 of duality, it can also be used to represent a pair of 20 orthogonal magnetic loop antennas.

Beginning from principles of the patterns of individual electrically short loops, equations governing the tandem . 23 performance can be derived and the conditions under which

steerable figure-eight responses can be determined. Thus, it
 can be shown that the square of the magnitude of the filed
 pattern of antenna system 12 is:

4

$$\left|H_{tot}\right|^{2} = K \cdot \left|\sin\theta\cos\alpha - \cos\theta + j\sin\theta\sin\alpha\right|^{2} = K(1 - \sin 2\theta\cos\alpha), \qquad (1)$$

6

7 where:

Φ is replaced with 90° because we are concerned with field
9 patterns in the y-z plane,

10 α is the phase difference of the current feeds for the 11 antennas wherein the feed currents are of equal magnitude, and 12 K is a constant.

13 In the event that $\alpha = 0^{\circ}$ (i.e., the two antennas are fed in 14 phase), the resulting pattern is a figure eight, rotated so that its null axis is aligned with 45 on the polar scale. However, 15 16 as α is increased (the antennas now being fed out of phase), the 17 null begins to vanish, leaving a peanut-shaped pattern. It can 18 also be shown that the net power varies only by approximately 19 1.5 dB over its domain for an 80 phase shift. The exact amount 20 of tolerance allowed on α , then, would depend on the amount of loss that can be tolerated due to the depth of the null in the 21

1 pattern. If high loss can be tolerated, a looser tolerance on α 2 is allowed.

For figure-eight pattern generation, consider now the more general case of two arbitrary antenna feed currents with an arbitrary phase difference between them. Suppose the currents are:

$$7 I_z = Ae^{j\alpha}, I_y = B. (2)$$

8 It can then be shown that the resulting total magnetic 9 field produced by this excitation will be

10

11
$$\vec{H}_{tot} = j\beta \Delta L \frac{e^{-j\beta r}}{4\pi r} \left(A e^{j\alpha} \sin\theta - B \cos\theta \right) \hat{\phi} , \qquad (3)$$

- 12
- 13 where:

14 ΔL is the length of the antennas, and

15 β is the wavenumber in the medium that the fields propagate 16 through.

To determine the farfield power pattern, the magnitude mustbe computed:

20
$$|H_{tot}|^{2} = \left| j\beta \Delta L \frac{e^{-j\beta r}}{4\pi r} \left(Ae^{j\alpha} \sin\theta - B\cos\theta \right)^{2} = K \cdot \left| A \left(\cos\alpha + j\sin\alpha \right) \sin\theta - B\cos\theta \right|^{2}.$$
(4)

Again the term K represents a term that is not dependent on
 the polar coordinate θ and does not need to be considered in
 determining the shape of the power pattern. Performing some
 simplifications, we find that this expression reduces to

5

$$\left|H_{tot}\right|^{2} = A^{2} \sin^{2} \theta + B^{2} \cos^{2} \theta - AB \cos \alpha \sin 2\theta.$$

(5)

7

6

8 To determine the possible ways of using the above equation 9 to give steerable figure eight.

10 If we make the substitution $X = A \sin \theta$, $Y = B \cos \theta$, the 11 power pattern reduces to

12

13
$$|H_{tot}|^2 = X^2 + Y^2 - 2XY \cos \alpha.$$
 (6)

14

This is a form recognizable as the Law of Cosines from 15 trigonometry. Using this knowledge, we can draw some 16 conclusions about the possible existence of nulls in the 17 18 pattern. If desired, we can visualize the power pattern's relationship to X and Y as a triangle where X and Y are adjacent 19 legs with angle α between the legs and $\left|H_{\iota o t}\right|^2$ is the leg opposite 20 angle α . Clearly, if $\alpha = 90^\circ$, then the triangle is a right 21 22 triangle, with the magnitude of the power patter as its

hypotenuse. Such a triangle cannot, according to the 1 Pythagorean Theorem, have a zero length hypotenuse unless both 2 legs of the triangle are of zero length. Therefore, it is not 3 possible to generate a pattern containing nulls at all with a 4 90 phase shift between the antennas. It is also apparent that 5 for angles $\alpha \neq 0^{\circ}$, it is not possible for a null to occur unless, 6 of course, both X and Y are zero. This, however, is an 7 impossibility, since X depends on sin θ , and Y depends on cos θ , 8 9 and these functions are never zero simultaneously.

10 To form a true null in the pattern, then, it is necessary 11 that $\alpha = 0^{\circ}$ (i.e., the two signals must be in phase). For in-12 phase currents, then the power pattern is given by

13

14
$$|H_{tot}|^2 = A^2 \sin^2 \theta + B^2 \cos^2 \theta - 2AB \sin \theta \cos \theta = (A \sin \theta - B \cos \theta)^2.$$
(7)

15

16 To locate the nulls in the pattern, we solve for angles θ 17 that render the equation zero:

18

19
$$A\sin\theta - B\cos\theta = 0 \Rightarrow \tan\theta = \frac{B}{A} \Rightarrow \theta = \arctan\left(\frac{B}{A}\right) \left(-90 \le \theta \le +90\right).$$
 (8)

20 We see here, then, that a single null axis exists for the 21 pattern and that its position is wholly dependent on the

magnitudes of the currents feeding the antennas. By appropriate 1 choices of A and B, then, we can steer the null in the pattern 2 to any desired location. From the point of view of 3 implementation, it would be desirable to remove the arctangent 4 . from the computation. This is accomplished by choosing A and B 5 such that $A = \cos \gamma$, $B = \sin \gamma$. In this way, the argument of the 6 arctangent function is the tangent of the angle y, the angle to 7 which the null is to be steered. The resulting power pattern 8 will be given by: 9

10

$$\left|H_{tot}\right|^{2} = \left(\sin\theta\cos\gamma - \cos\theta\sin\gamma\right)^{2} = \sin^{2}(\theta - \gamma).$$
(9)

12

13 This pattern may be recognized as that of a standard 14 figure-eight antenna pattern rotated by γ degrees. The null in 15 the pattern remains well defined for low phase differences; the 16 exact tolerance on α will depend on the requirement for the null 17 depth of the figure-eight pattern.

In a real network, it will not always be possible to create a situation where the two signals from the antennas possess the exact magnitudes and phases to produce a perfect omni or figureeight pattern. It is of value to see what effect the errors in

magnitude and phase will have on the resulting patterns for the
 figure-eight antenna.

3 The location of the minimum in the pattern can be 4 determined where the value of θ satisfies

5

$$6 \qquad \qquad \frac{\partial}{\partial \theta} |H_{tot}|^2 = 0 \Longrightarrow \frac{\partial}{\partial \theta} \left(A^2 \sin^2 \theta + B^2 \cos^2 \theta - AB \cos \alpha \sin 2\theta \right) = 0. \tag{10}$$

7

8 By working out the partial derivative, it can be shown that 9 for positive values of A and B the pattern minimum occurs for 10

11
$$\theta_{\min} = \frac{1}{2} \arctan\left(\frac{2AB\cos\alpha}{A^2 - B^2}\right) + \frac{m\pi}{2} \qquad m = 0, 2, 4, \dots$$
 (11)

12

13 It can also be shown that the pattern maximum occurs 90
14 away from the pattern minimum:

15

16
$$\theta_{\max} = \frac{1}{2} \arctan\left(\frac{2AB\cos\alpha}{A^2 - B^2}\right) + \frac{n\pi}{2}$$
 $(n = 1, 3, 5, ...) = \theta_{\min} + \frac{\pi}{2}$ (12)

17

18 The min-to-max ratio can now be determined by substituting 19 the above values of θ into the expression for the general power 20 pattern. It can be shown that the resulting M is given by:

$$M = \frac{A^2 \sin^2 \theta_{\min} + B^2 \cos^2 \theta_{\min} - AB \cos \alpha \sin 2\theta_{\min}}{A^2 \cos^2 \theta_{\min} + B^2 \sin^2 \theta_{\min} + AB \cos \alpha \sin 2\theta_{\min}}.$$
(13)

2

From these equations, it can be seen that errors in the magnitudes and phase of the currents on the antennas can lead to non-ideal performance when the signals are combined to produce the figure-eight pattern. For instance, a non-zero phase difference α will cause a position error in the location of the null in a figure-eight pattern.

FIG. 3 shows a functional block diagram of system 10A to 9 combine two orthogonal patterns into omni and/or steerable 10 figure-eight patterns. The two antenna inputs 18 and 20 are 11 referred to as the F/A (fore/aft) and ATH (athwart) antennas, 12 respectively. Variable gain amplifiers 22 and 24 have a gain of 13 $\cos \gamma$ and $\sin \gamma$, respectfully, where the angle γ input to steer 14 to is indicated at 26. Summation output 32 produces a steerable 15 figure-eight output. 16

17 The details of an electronic circuit to perform the 18 functions shown in FIG. 3 for a steerable figure-eight output 19 are indicated in FIG. 2. This circuit is specifically tailored 20 to VLF/LF radio waves. One of ordinary skill in the art could 21 devise a similar circuit for higher frequencies using monolithic 22 components. In FIG. 2, F/A antenna input is preferably provided

by a 50 Ω cable to terminals F/A Hi and F/A Lo, designated as 34 1 and 36, respectively. The F/A antenna input signal is then 2 isolated by transformer 38. Likewise, the ATH antenna input is 3 provided to terminals ATH Hi and ATH Lo, designated as 40 and 4 42, respectively, by a 50 Ω cable and is isolated by transformer 5 44. Capacitors 46 and 48 provide filtering on the input 6 signals. Inversion of the ATH channel, if needed depending on 7 the desired quadrant of the antenna pattern, is provided by 8 relay 50 in response to an invert signal on line 60. Relay 50 9 selectively reverses the polarity of the ATH signal before the 10 ATH signal arrives at the ATH variable gain amplifier which is 11 comprised of digital-to-analog component (DAC) 52 and amplifier 12 54 which components may preferably be connected together in a 13 feedback loop to form a variable gain amplifier. The F/A signal 14 is applied to F/A variable gain amplifier which is comprised of 15 DAC 56 and amplifier 58 preferably connected together in a 16 feedback loop to form a variable gain amplifier. 17

The gain settings for the F/A variable gain amplifier and the ATH variable gain amplifier are supplied by F/A steering input 62 and ATH steering input 64 by software controlling the system. Thus, inputs 62 and 64 may typically each comprise a data bus. The F/A gain setting is stored in a register within DAC 56 and the ATH gain setting is stored in a register with DAC

1 52. The software for antenna goniometer 10 computes, from a 2 given null steering angle φ , the required gains of the F/A and 3 ATH channels: sin φ and cos φ , respectively. Inversion of the 4 ATH signal by inverter 50 is provided to allow steering into the 5 2nd and 4th quadrants of the plane of the antenna pattern.

6 Trimmer potentiometers 66 and 68 provide zero-offset 7 adjustment for the variable gain stages. Capacitors 70 and 72 shunt any out-of-band high frequency (HF) interference from the 8 9 signal lines. The scaled signals are then fed to summing 10 amplifier or processor 74 formed by amplifier 76 and resistors 11 78, 80, and 82. Variable resistor 82 provides a manual gain 12 adjustment. The summation of signals is provided at junction 13 84.

Buffer amplifier 86, together with resistor 88 and transformer 90 are preferably utilized to provide a 50Ω drive stage to drive the balanced output lines 92 and 94.

17 Thus, the operation of circuit 10 is as follows and is in 18 accord with the mathematical theory of steerable antenna 19 patterns described hereinbefore. The signals from the 20 orthogonal antennas denoted herein by F/A (Fore/Aft) and ATH 21 (Athwart) enter the system as described hereinbefore at inputs 22 34, 36 and 40, 42, respectively. The signals are directed to 23 respective variable gain amplifiers. The F/A variable gain

amplifier is comprised of DAC 56 and amplifier 58. The ATH 1 variable gain amplifier is comprised of DAC 52 and amplifier 54. 2 The gain settings for the variable gain amplifiers are 3 preferably stored in a register in the respective DAC 56 and DAC 4 The gain settings come from the controlling software. 5 52. The software has a F/A steering link input at 62 and an ATH steering 6 link at 64. Thus, links 62 and 64 preferably comprise suitable 7 digital data busses that are used for inputting the gain 8 settings into the respective registers within DAC 52 and DAC 56. 9 10 Software is preferably utilized to compute the required gain settings from a given null steering angle and input the gain 11 settings whenever desired. The gain settings may be relatively 12 constant or may change to create some desired pattern, such as a 13 14 rotating figure-eight pattern, with a desired timing such as a desired number of rotations per minute. An inversion of the ATH 15 signal is provided by inverter 50 to allow steering into the 2nd 16 17 and 4th quadrants of the plane of the antenna pattern. The 18 resultant signals from the variable gain amplifier are added by 19 a summing amplifier and presented to the output at 92 and 94.

20 While a particular circuit for a presently preferred 21 embodiment has been described, it will be understood that the 22 invention is not limited to this particular circuit and could be 23 implemented in other ways.

It will be appreciated by those skilled in the art that the 1 invention can also be implemented using a suitable programmed 2 3 general purpose computer and/or special purpose hardware, with program routines or logical circuit sets performing as 4 processors. Program routines, electronic circuits, or logical 5 6 circuit sets may also be referred to as processors or the like. It will be understood that many additional changes in the 7 details, materials, steps and arrangement of parts, which have 8 9 been herein described and illustrated in order to explain the nature of the invention, may be made by those skilled in the art 10 within the principle and scope of the invention as expressed in 11 12 the appended claims.

1 Attorney Docket No. 82646

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4 5

DIGITAL ANTENNA GONIOMETER AND METHOD

ABSTRACT OF THE DISCLOSURE

A digital goniometer and steering method are provided which 6 may be used, in a preferred embodiment, for producing a figure-7 eight antenna reception pattern oriented at any selectable angle. 8 The digital goniometer works with the signals from a pair of 9 orthogonal antenna loops. In a presently preferred embodiment a 10 11 first register and a second register are provided for a respective first variable gain amplifier and a second variable gain amplifier. 12 13 The first and second registers receive from software a digital value related to a sine function and cosine function of the 14 selectable angle. The first and second registers are utilised to 15 set the gain of the first and second variable gain amplifiers. 16 The 17 outputs of the first and second variable gain amplifiers are added to produce the reception signal. 18



