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# DETERMINATION OF DESIRABLE LENGTHS OF

Z- AND CHANNEL-SECTION COLUMNS

FOR LOCAL-INSTABILITY TESTS

By George J. Heimerl and J. Albert Roy

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NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS



DETERMINATION OF DESIRABLE LENGTHS OF

Z- AND CHANNFL-SECTION COLUMNS

# FOR LOCAL-INSTABILITY TESTS

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## SUMMARY

Local-instability tests of 24S-T aluminum-alloy formed Z- and channel-section columns were made in order to determine a length of test specimen that would avoid the increased strength associated with short lengths and also permit the occurrence of a convenient buckling pattern. The effect of column length on the critical compressive stress, on the average stress at maximum load, and on the number of half-waves of the buckling pattern is shown. A buckling pattern of three half-waves is indicated as desirable for test purposes. A curve is presented from which may be determined the lengths of Z- or channel-section columns that give a buckling pattern of three half-waves. When the strength for local instability is very high, a reduction in the length indicated by the curve may be necessary to prevent column failure. In order to avoid the increased strength associated with short lengths, a ratio of length to web width above 3.5 should be used.

## INTRODUCTION

In local-instability tests of Z- and channel-section columns, suitable specimen lengths should be determined. As the flanges and webs of such columns may be considered plates with various kinds of edge supports, the local instability of these columns becomes a plate-buckling problem. When a plate is long, the critical compressive stress tends to be independent of length; whereas, if the plate is very short, the stress increases appreciably. (See fig. 6 of reference 1.) For an investigation of columns that develop local instability, therefore, the soccimens should be made long enough to avoid an appreciable increase in stress and yet not long enough to result in column failure. In order to determine lengths of columns that meet these requirements, tests were made of formed Z- and channel-section columns of various lengths. This report presents the test results and a curve for determining desirable column lengths for test purposes.

# SYMBOLS

b<sub>P</sub> width of flange, inches

b<sub>w</sub> width of web, inches

L length, inches

t thickness of web or flange, inches

E modulus of elasticity, ksi

 $\sigma_{\rm op}^{}$  critical compressive stress, ksi

 $\sigma_{\rm max}$  everage stress at maximum load, ksi

 $\sigma_{
m ev}^{}$  compressive yield stress, [kci

## SPECIMENS.

Formed Z- and channel-section columns were made from 2hS-T aluminum alloy with the grain of the material parallel to the length of the column; one sheet of material was used for each type of column, and 73 columns of each type were tested. The ends of the specimens were ground flat, parallel, and at right engles to the length of the column. Figure 1 shows the nominal dimensions of the three cross sections used for the Z- and channelsection columns. The measured dimensions of the columns and the test results are given in table 1. For each cross section, the ratio of length to web width was varied from about 1 to about 10. Stress-strain tests of the material were made with single-thickness specimens in a roller-type compression fixture similar to that shown in figure 2 of reference 2. Compressive stress-strain curves are shown in figure 2. The values of the compressive yield stress, determined by the C.2-percent-offset method, and of the modulus of elasticity are given in table 2.

## METHOD OF TESTING

The column tests were made in a 300,000-poundcapacity compression testing machine that is accurate within three-quarters of 1 percent for the range of load used in the tests.

A Z-section column under test is shown in figure 3. The displacement of pointers, supported by extension arms attached to the flanges of the columns, was measured by the optical micrometers that can be seen in figure 3. The critical compressive stress was obtained from stressdistortion curves in the manner described and illustrated in reference 3. In this method, the critical stress is determined as the point near the top of the knee of the stress-distortion curve where a marked increase in distortion first occurs with small increase in stress.

## RESULTS AND DISCUSSION

The variation of  $\sigma_{cr}$  and  $\sigma_{max}$  with  $L/b_W$  for each of the different types of column tested is presented in figure 4. Columns having  $\frac{b_W}{t} = 2h$  and  $\frac{b_F}{b_W} = 0.5$ developed bending failure for  $\frac{L}{b_W} > 7$ . A definite rise in critical and maximum stresses when the columns become very short is shown by these curves. For all except the very short columns, however, the curves are relatively level. The number of half-vaves of the buckling pattern that occurred in each case is also indicated in figure 4.

It has been found desirable for test purposes to make the column length such that an odd number of halfwaves develops, because of the convenience in measuring cross-sectional distortion at the center of the column. Economy of material and the possibility of banding failure if the column is long lead to a choice of lengths such that the columns will develop the least number of halfwaves and still avoid an appreciable increase in stress due to the effect of short lengths. These considerations, together with the test results shown in figure 4, indicate that a buckling pattern of three half-waves is the one most desirable for investigations of local instability of columns.

The length of the half-wave developed when local instability occurs varies, for a given web width, with the cross-sectional ratio  $b_{\rm H}/b_{\rm W}$ . The number of half-waves then depends on the ratio of length to web width  $L/b_{\rm W}$ . By showing the number of half-waves that occur for given values of  $b_{\rm F}/b_{\rm W}$  and  $L/b_{\rm W}$  as illustrated in figure 5, curves may be drawn that show the relationship between  $b_{\rm F}/b_{\rm W}$  and  $L/b_{\rm W}$  required to obtain any desired number of half-waves. In order to give proper weight to the tost results shown in figure 5, the number of tests for which each number of half-waves occurred is indicated.

A recommended curve is drawn in figure 5 to indicate the proportions of either a Z- or a channel-section column required to develop a buckling pattern of three halfwaves, which is desirable for test furposes. This curve can be used directly for selecting specimen lengths in many cases. In cases in which the strength for local instability is high (low values of  $b_W/t$ ), however, specimen lengths selected according to the recommended curve may not be short enough to prevent bending failure. It is therefore necessary to check the column strength of the specimens selected and, in some cases, to shorten the specimens. In any case, figure h shows that, in order to avoid an increase in strength due to the effect of very short lengths, the value of  $L/b_W$  used should be above 3.5.

#### CONCLUSIONS

For local-instability tests of Z- and channelsection columns, the specimens should be just long enough to avoid the increased strength associated with short lengths but of such length that a buckling pattern convenient for test purposes occure. A buckling pattern of three half-waves meets these requirements; the proper length for this condition may be obtained from a curve

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based on tests. When the strength for local instability is very high, a reduction in this length may be necessary to prevent column failure. In order to avoid the increased strength associated with short lengths, a ratio of length to web width above 3.5 should be used.

Langley Memorial Aeronautical Laboratory National Advisory Jonmittee for Aeronautics Langley Field, Va., August 10, 1944

## REFE RENCES

- Lundquist, Eugene F., and Stowell, Elbridge Z.: Critical Compressive Stress for Flat Rectangular Plates Supported along All Edges and Elastically Restrained against Potation along the Unloaded Edges. NACA Rep. No. 733, 1942.
- 2. Paul, D. A., Howell, F. M., and Grieshaber, H. E.: Comparison of Stress-Strain Curves Obtained by Single-Thickness and Pack Methods. NACA TN No. 819, 1941.
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Specimen	t (in.)	<b>b</b> w (in.)	b <sub>F</sub> (in.)	(in.)	L bw	bw t	b <sub>F</sub> b₩	σ <sub>cr</sub> (ksi)	σ <sub>max</sub> (ksi)	Number of half- waves
Z-section column; section 1										
1a 1b 2a 3a 3b 3c 4b 4c 5b 5c 6a 6b 6c 7	$\begin{array}{c} 0.106\\ .106\\ .105\\ .105\\ .106$	2.2.2.2.2.5.55555555555555555555555555	1.30 1.30 1.30 1.30 1.30 1.30 1.30 1.30	2.79 2.79 5.355 7.60 10.30 10.30 12.92 12.93 15.50 15.54 15.54 17.60	1.10 1.10 2.12 3.11 2.99 3.07 4.07 4.05 5.07 5.08 5.07 6.09 6.19 6.11 6.97	24.00 24.059 24.059 24.0000 24.0000 24.0000 24.0000 24.0000 24.0000 24.0000 24.0000 24.0000 24.0000 24.00000 24.0000000000	0.51 .51 .51 .51 .51 .51 .51 .51 .51 .51	47.4 445.2 43.9 412.1 43.9 42.1 40.0 79.8 49.8 8 49.8 8 39.8 8 39.8 8 39.8 8 39.8 39.8 39.	50.22 50.22 44.13 143.51 443.51 443.51 443.51 443.2 43.2 43.2 43.2 43.2 43.2 43.2 43.	1 1 1 2 2 2 2 2 3 3 3 4 4 4 5
Z-section column; section 2										
8 a 8 b 8 c 9 a 9 c 10 c	$\begin{array}{c} 0.105\\ .105\\ .105\\ .105\\ .106$	2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.	2.2.2.2.5.58 5.5.58 2.2.2.2.5.56 2.2.2.2.2.5.5.55 2.5.5.58 2.2.2.2.2.2.2.2.5.55 2.555 2.555 2.555 2.555 2.555 2.555 2.555 2.555 2.5555 2.5555 2.55555	2.72 2.68 2.68 5.33 5.77 6.725 10.38 10.38 12.95 15.55 15.55 15.55 160.08 20.62 20.22 23.22 25.65 14 2.06 20.62 20.22 25.65 25.74	1.07 1.05 1.06 2.08 2.03 2.64 4.06 4.06 5.07 5.10 6.11 6.11 6.11 6.11 6.11 8.10 8.11 9.13 9.13 9.13 9.13 10.08 10.11	141114100201 2411141400201 2411141400201 2411141400201 2411141400201 2411141400201 2411141400201 241114100005598 241100005598 2411141400201 24111400201 24111400201 24111400201 24111400201 24111400201 2411140020000000000	$\begin{array}{c} 1.02\\ 1.01\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.01\\$	08208655312519296617721419265555 21555165551929661772141992655555 16655514555145514551415514 154415555	8.0.1.0.9.2.56.3.2.4.2.7.9.5.5.1.4.8.7.6.5.3.9 2.0.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2	1 1 1 1 1 1 1 1 1 1 1 1 1 2 2 2 2 2 2 2
Z-section column; section 3										
18a 18b 19 20a 20b 20c 21a 21b 21c 22b 22c 23a 23b 23c 25a 25a 25a 25a 25a 25a 25a 25a	$\begin{array}{c} 0.\ 105\\ .\ 105\\ .\ 105\\ .\ 105\\ .\ 106$	1.40 $1.40$ $1.41$ $1.42$ $1.42$ $1.42$ $1.42$ $1.42$ $1.42$ $1.42$ $1.44$ $1.44$ $1.44$ $1.44$ $1.44$	$1.48 \\ 1.48 \\ 1.48 \\ 1.48 \\ 1.45 \\ $	1.1.777903003223444580042242288766 4.4.5.5.5.7.7.7888.00.0.121.5.5.9.9.3323744458004242288766	$\begin{array}{c} \textbf{1.25}\\ \textbf{1.256}\\ \textbf{3.115}\\ \textbf{5.111}\\ \textbf{5.111}\\ \textbf{5.111}\\ \textbf{5.111}\\ \textbf{4.107}\\ \textbf{7.070}\\ \textbf{6.0999}\\ \textbf{9.993}\\ \textbf{9.991}\\ \textbf{8.991}\\ \textbf{8.991}\\ \textbf{9.991}\\ \textbf{9.991}\\ \textbf{9.995} \end{array}$	299228 133.5.38920 133.5.38920 133.5.38920 133.5.44533 133.5.56622 133.5.6622 133.5.665 133.5.5660 133.5.5660 133.5.5600 133.5.5600	$\begin{array}{c} 1.06\\ 1.06\\ 1.06\\ 1.03\\ 1.05\\ 1.04\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 1.02\\ 98\\ 98\\ 98\\ 98\\ 98\\ 98\\ 98\\ 99\\ 98\\ 99\\ 99$	899209612569224360615687697 44426261325692244360615687699 44389112213600115687699 4440990	734 184756826963821867559034 55514645553443344335512444444 55514661756826963821867559034	1 1 1 1 1 2 2 2 2 3 2 3 3 3 3 3 4 4 4 4 4 4 4 4 4

TABLE 1 MEASURED DIMENSIONS OF FORMED SPECIMENS AND TEST RESULTS

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Specimen	t (in.)	b. (in.)	<sup>b</sup> <sub>F</sub> (in.)	(in.)	L bw	b <b>y</b> t	₽ ₽	σ <sub>er</sub> (kei)	C <sub>max</sub> (ksi)	Number of half- waves
Channel-section column; section 1										
111222255544555556667777	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	ઌૻૺૡૺૼઽ૾ૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢૢ	1.26 1.26 1.27 1.26 1.26 1.28 1.28 1.28 1.28 1.26 1.26 1.26 1.26 1.26 1.26 1.26 1.26	2.2.2.5.5.7.7.7.0.0.7.3.1 10.0.3.5.00.7.3.1 10.0.3.5.00.7.3.1 11.1.0.0.5.1 10.0.5.10	1.09 1.10 1.05 2.11 2.10 3.06 3.11 4.07 4.06 5.11 5.11 5.08 6.00 6.10 7.09 7.11	2858851885188851888518885188851888514885148851488514885148851488514885148851488514885148851488514885148514		4452.176479609994400.2079977	999444674454949444949494995 999444647445494949494949494995 918676881464079449494995995	111111500000000000000000000000000000000
Channel-ssotion column; eaction 2										
8a 8b 9a 9b 10b 10c 11a 11b 12a 13b 13c 13b 13c 15c 15c 15c 15c 15c 15c 15c 15	0.1055 1055 1055 1055 1055 1056 1056 1056		๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛	2.90 2.57 2.80 5.5381 5.5381 10.388 10.388 12.986 10.388 12.986 10.388 12.9866 12.986 12.9866 12.9866 12.986 12.9866 12.9866 12.	1.10 1.112 1.112 1.112 1.112 1.00 1.	1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1	$\begin{array}{c} 1.00\\ .99\\ 1.00\\ 1.00\\ 1.00\\ 1.01\\ .99\\ .99\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ 1.00\\ .99\\ .00\\ .99\\ .00\\ .99\\ .00\\ .99\\ .00\\ .99\\ .00\\ .99\\ .00\\ .99\\ .00\\ .99\\ .00\\ .00$	1 3990 50 538 757946 8760 592470 76 5	75559854690070554490058010655610	111111112222222355555555555555555555555
Channel-section column; section 3										
18 19a 19b 20a 21a 21b 21c 22c 23b 23c 23b 23c 24a 24b 25a 25b 25c 25b 25c 26b	0.106 .105 .105 .105 .105 .106 .106 .106 .106 .106 .106 .106 .106	1.41 $1.44$ $1.443$ $1.443$ $1.445$ $1.445$ $1.445$ $1.445$ $1.445$ $1.445$ $1.445$ $1.445$ $1.444$ $1.445$ $1.440$ $1.443$		1.64 2.83 3.098 4.439 5.598 5.593 7.7326 8.520 10.067 11.650 11.650 11.555 11.555 22.92	1.16 1.971 2.08 3.08 4.01 4.01 4.01 4.05 5.05 5.01 9.05 7.05 7.07 2.28 8.23 8.00 8.23 8.00 1.25 7.00 2.00 1.05 7.00 2.00 2.00 2.00 2.00 2.00 2.00 2.00	$13.29 \\ 13.67 \\ 13.61 \\ 13.65 \\ 13.65 \\ 13.4$	1.00 .999 .98 .98 .98 .99 .98 1.03 .98 .99 1.03 1.00 1.00 1.00 .999 .972 1.01 1.00	48.1.1.12.5.2.52.52.54.55.54.4.9.50.67 9.3.4.2.5.2.52.52.55.55.4.9.50.67 9.3.4.2.5.2.52.55.55.55.4.9.50.67	6 3 3 4 9 37 - 3 26 7 6 1 28 30 20 1 37 1 54 9 8 7 5 5 5 4 3 3 5 5 5 3 8 3 5 3 7 1 0 2 6 9 9 8 7 5 5 5 4 3 5 5 5 5 3 8 3 5 3 7 1 0 2 7 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	1111112222255555544444

# TABLE 1 - Concluded

MEASURED DIMENSIONS - Concluded

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# TAPLE 2

# COMPRESSIVE PEOPERTIES OF MATERIAL

	With gr	rain	Cross grain			
Coupon	E (ksi)	σ <sub>cv</sub> (ks <u>i</u> )	E (ksi)	σ <sub>cy</sub> (ksi)		
Z-section	10,700	44.8	10,600	49.7		
Channel section	10,600	44.2	13,600	49.2		

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Section 3

Z-section

Channel section NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS



Fig. 1







Figure 3.- Local instability of a Z-section column.

Fig. 3



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Z - section

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Fig. 4

Fig. 5



produced by various proportions of Z- and channel-section columns of 24 S-T aluminum alloy.

<ul> <li>ITTLE: Determination of Desirable Lengths of Z- and Channel-Section Columns for Local- Instability Tests</li> <li>AUTHOR(S): Heimerl, George; Roy, Albert J.</li> <li>ORIGINATING AGENCY: National Advisory Committee for Aeronautics, Washington, D. C.</li> <li>PUBLISHED BY: (Same)</li> </ul>								HO.
Date 244	DOC. CLASS.	COUNTRY 11 S	LANGUA	PAGES	ILLUSTRATIONS	liams maphs		
APSTRACT.	Unclass,	<u> </u>	Eng.	13	photos, d	uagrs, graphs	· · · ·	
For local instability tests of 24 S-T aluminum-alloy formed Z- and channel-section columns the specimens should be just long enough to avoid the increased strength associated with short lengths but of such length that a buckling pattern convenient for test purposes occurs. A buckling pattern of three half-waves meets these requirements. The proper length for this condition may be obtained from a curve based on tests. In order to avoid the increased strength associated with short lengths, a ratio of length to web width above 3.5 should be used.								
SECTION: Stru	v and Analysi	s (7) s	SUBJECT FIEADINGS:					
Methods (2)				Stress analysis (90800); Strength of materials (90750); Columns-Stability (23595.3)				
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Air Documents Air	Divisian, Intollig Materiol Comma	once Dopartment nd	AIR	TECHNICAL IN	DEX	Wright-Patterson Air Dayton, Oh	Force Base lia	-1