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DEVELOPMENT OF ARMY HIGH-ENERGY FUEL FOR DIESEL/ TURBINE-POWERED SURFACE EQUIPMENT – PHASE II

INTERIM REPORT AFLRL No. 147

By

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U.S. Army Fuels and Lubricants Research Laboratory Southwest Research Institute San Antonio, Texas

Under Contract to

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20. ABSTRACT (Cont'd)

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fuels and fuel components are described. The fuel components included both liquids and solids (at room temperature). The blending of the various components and the characterization of the resulting fuels are outlined. Solubility studies were done on some of the solid components to assist in obtaining the resulting fuel in the more desirable liquid state. Those solid components, such as anthracene, that were insoluble in the tests conducted but judged to have good high-energy potential, were studied as slurries.)

Catalyzed "carbon black" was investigated. Settling studies with carbonaceous fuels included the effect of temperature upon the stability, the stability of FRF-carbon slurries, and the refinement of the previously reported "Freezing Tube Technique" for measuring stability.

⁴Impact dispersion tests were conducted to study the fire-safety characteristics of energy-augmented fuels. The screening of candidate fuels with the Petter engine is described in detail. Other engine studies with the CLR engine are also included.

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FOREWORD

The work reported herein was conducted at the U.S. Army Fuels and Lubricants Research Laboratory (AFLRL), Southwest Research Institute, San Antonio, TX under Contracts DAAK70-80-C-0001 and DAAK70-82-C-0001. The work was funded by the U.S. Army Mobility Equipment Research and Development Command (MERADCOM), Ft. Belvoir, VA. Contracting Officer's representative was Mr. F.W. Schaekel, Fuels and Lubricants Division, Energy and Water Resources Laboratory (DRDME-GL).



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I. INTRODUCTION

The Army Scientific Advisory Panel, in its Summer Study 76 $(\underline{1})^*$, recommended a research program on high-energy fuels, i.e., fuels that could increase the payload capabilities and/or extend the operational range of both its tactical and combat vehicles. The development of such a fuel or fuels would need to be conventional enough that they could be used in present and future Army vehicles, yet novel enough that they contain the high-energy (per unit volume) requirement necessary to fulfill the need.

The Air Force and Navy are also interested in high-energy (high-density) fuel development. The Air Force at its Aero Propulsion Laboratory at Wright-Patterson Air Force Base has been actively evaluating hydrocarbon blends that are potential performance improvers for its air-breathing type missile systems. This class of fuel would be exceptionally good for use in the strategic cruise missile whose maximum effectiveness depends on maximum range, minimum time to reach the target, and minimum weight. Since small missiles, such as the cruise, are volume-limited, it is imperative that the fuel contains a high-energy content per unit volume. In general, an increase in the volumetric energy content of a fuel will result in the range of the missile being increased by a like percentage. Although several ways are available to increase the energy content of a given quantity of fuel, the most applicable for missile use is to increase the fuel's density. Research on high-density, high-energy liquid hydrocarbon fuels by the Air Force and Navy continues. This work is being closely monitored by the U.S. Army. A few of these fuels that may be of interest to the Army are described later. The purpose of this program was recently reaffirmed and summarized in a letter from U.S. Army Mobility Equipment Research and Development Command (USAMERADCOM) to Army Fuels and Lubricants Research Laboratory (AFLRL).(2) The letter stated that the purpose of the High-Energy Fuel Program is to provide the user with a fuel having the following characteristics:

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^{*}Underscored numbers in parentheses refer to the list of references given at the end of this report.

 Increases range by 10 percent without any increase in fuel tank volume;

- Does not require any engine modification;
- Is in liquid form for use in existing equipment (two-phase fuel acceptable);
- Has no detrimental effect on the engine;
- Is able to use the existing fuel-handling and transportation equipment;
- Displays no unusual toxicology problems (compared to existing petroleum fuels);
- Considers cost when the above-mentioned requirements are achieved;
- Is available for user acceptance test by FY83;
- Is not considered as a booster for the FRF only, but should perform by itself, as specified.

An interim report, $(\underline{3})$ released in October 1979, covered the progress made on the Army High-Energy Fuels Program from October 1977 until October 1979. The current report covers the period from October 1979 through September 1981.

II. APPROACH

The approach has been to identify those fuels and fuel components that have the potential of being high-energy fuel candidates or components for preparing such fuels. A series of screening processes were used to single out these fuels.

The initial screening uses the volumetric heat of combustion (ΔHc) as the main criterion. The physical and chemical properties of the fuel and fuel components are determined in the laboratory whenever possible or values from the chemical literature are used. Density and percent hydrogen are determined on the fuel whenever feasible since they are used to convert the

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gravimetric determined (ASTM D 240-76) gross ΔHc to the net volumetric ΔHc value.

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The experimental gross gravimetric heat of combustion, as measured by ASTM Method D 240, is usually reported in British thermal units (Btu) per pound. If the hydrogen content of the sample is known, this value can be converted to the net gravimetric heat of combustion. Knowing the density of the sample, the volumetric heat of combustion can be obtained. In this report, this value will be expressed in Btu per gallon.

ASTM Method D 240 states that duplicate results by the same operator should be considered suspect if they differ by more than 55 Btu per pound and the results by two or more laboratories should be considered suspect if they differ by more than 175 Btu per pound.

In this report, as a matter of convenience in presenting and comparing the data, the last digit in the gravimetric heat of combustion value is adjusted to the nearest multiple of five. This can be done since the last digit is an order of magnitude less than the precision of the method.

Most of the fuels and fuel blends described in this report are liquids at room temperature. The selection was intentional, as liquid fuels are easier to handle and transport in existing fuel equipment and would undoubtedly run with less problems in diesel and turbine type engines than slurried fuels. However, promising slurried fuels were also tested. The second screening process uses a Petter engine as the screening tool. This engine, described in detail later, was adequate for this use. A highly instrumented CLR engine was used in a few of the screening tests, but because of the uncertainty of its availability when needed and its time-consuming operation, a simple screening device was needed. The Petter engine proved to be a suitable choice. This CLR engine will be used henceforth only in testing those fuels showing the greatest promise from the results of the first two screening tests. Only after passing this final screening with the CLR engine was the candidate fuel considered for the extensive engine tests necessary for qualifying it as a high-energy fuel.

Figure 1 illustrates the steps in the testing procedure for qualifying a high-energy fuel.



FIGURE 1. FLOW CHART OF TESTING PROCEDURE FOR EVALUATING HIGH-ENERGY FUEL

MERADCOM Commander, during a visit to AFLRL 13 March 1978, commented on the desirability of developing a high-energy fuel that was fire safe, and that the High-Energy Fuel (HEF) Program and the Fire-Resistant Fuel (PRF) Program should eventually merge.

Efforts have been made during the last 2 years to interface the HEF and FRF programs whenever possible, without diluting the goal of achieving a highenergy fuel that masts the characteristics described earlier.

III. TECHNICAL PROGRAM

A. Liquid High-Energy Components

Some of the liquid components described below and listed in Table 1 could

Component	Density 15.56°C (60°F) (g/ml)	∆Hc (net) Btu/gal.	∆Hc (net) <u>Btu/1b</u>	_\$/Gal
RJ-5	1.089	160,390	17,735	86.00(a)
JP-10	0.940	144,615	18,435	15.15(a)
rj-6m	1.001	150,070	17,970	40.00(a)
Tetralin				
(Tetrahydrouaphthalene)	0.972	140,990	17,580	13.90(b)
Furfural Extract SRC-II	0.987	143,415	17,565	1.25(c)
(a) 2.9:1 middle distillate				_
to heavy distillate	0.999	137,895	16,540	0.62 to 0.69(d)
(b) middle distillate	0.984	134,755	16,410	0.61 to 0.69(d)
Polycyclic Aromatic				
Blending Stock	1.017	140,340	16,340	1.38(e)
No. 5 Burner Oil	0.917	137,700	17,995	1.90 to 1.93 (f)
No. 6 Fuel 011	0.978	141,575	17,350	0.80 to 0.82
DF-2 (Cat 1-H)	0.8504	132,365	18,665	1.69(h)
DF-2 (Typical)	0.8484	130,615	18,450	1.00

TABLE 1. LIQUID HIGH-ENERGY COMPONENTS

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(a) Estimated price in quantities of 5000/6000 gal. (Ashland Chemical Co.)
(b) Estimated price in 40,000 gal. quantities (DuPont)

(c) Estimated price--Ashland Chemical Co.

(d) Wholesale net back to the plant: Bulk shipment. Excludes transportation, tax, handling, sales margin, distribution costs.(4)

- (e) Tank car; tank truck FOB W. VA.
- (f) Truck transport, FOB Chicago
- (g) Truck transport, FOB Baltimore (0.5% sulfur)
- (h) 8,000-10,000 gallons

possibly be used as fuels without further modification or blending with other materials; some may require a small amount of cetane improver; and some may be too costly to use neat but are desirable as a component because of some beneficial property, such as a very high volumetric heat content, viscosity improver, etc. 1. RJ-5

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This synthetic liquid hydrocarbon fuel was described in an earlier report. (3) It was one of the first high-density fuels of the polycondensed xy-loparaffic type studied and is currently being produced by Ashland Chemical Company. The typical properties of this fuel, as supplied by the manufacturer, are listed in Table 2. RJ-5 has the highest volumetric heat JUNE RUN

TABLE 2. TYPICAL PROPERTIES OF RJ-5

Specific Gravity, 15.56°C (60°F)	1.06 (min)
Existent Gum, mg/100 ml	7.0 (max)
Aromatics, vol%	1.0 (max)
Sulfur, total, wt%	0.06 (max)
Pour Point, *C (*F)	-28.9 (-20)
Net heat of combustion, Btu/1b	17,750
Net heat of combustion, Btu/gal.	160,000 (min)
Viscosity, cSt	
at -53.9°C (-65°F)	20,000 (max)
at -1.1°C (-30°F)	1,400 (max)
at 37.7°C (100°F)	15 (max)
Flash Point, °C (°F)	65.6 (150)

of combustion of any liquid component tested (~160,000 Btu/gal.), but is expensive. Impact dispersion tests (5) using fuel blends of RJ-5 with DF-2 and with FRF indicated that RJ-5 had fire-resistance properties under the conditions of the test. The result of these tests are described in Section III. E. of this report.

2. JP-10 (Exo-Tetrahydrodicyclopentadiene)

JP-10 was also described in Reference 3. The typical properties of JP-10 are shown in Table 3.

TABLE 3. TYPICAL PROPERTIES OF JP-10 Exo-tetrahydrodicyclopentadiene $(C_{10}H_{16})$ Molecular Weight 136.2 0.62 C:H Ratio Specific Gravity @ 15.6°C (60°F) 0.94 142,000 Heating Value, Btu/gal. Viscosity, cSt $@-40^{\circ}C$ (-104°F) 19 Freezing Point, °C (°F) -78.9 (-110) Flash Point, *C, (*F) 57.2 (135)

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3. <u>RJ-6M</u>

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This high-energy candidate is the latest to emerge. It has been tested in the laboratory and later screened using the Petter engine. RJ-6M was developed by Ashland Chemical Company, and its composition at present is proprietary. Table 4 lists the typical properties of this fuel as supplied

TABLE 4. PROPERTIES OF RJ-6M

Density @ 15.56°C (60°F)	1.001 g/ml (8.352 1b/gal.)
Net Heat of Combustion	
Btu/lb	17,968
Btu/gal.	150,069
Viscosity @	
$-17.8^{\circ}C$ (0°F)	68.5 cSt
-31.7°C (-25°F)	110 cSt
-34.4°C (-30°F)	240 cSt
-40.0°C (-104°F)	360 cSt
Flash Point, *C (*F)	65 (140)

by the manufacturer.* The AFLRL experimentally determined cetane numbers, with and without cetane improver, are listed in Table 5. This fuel is a good high-energy candidate.

*Ashland Chemical Co., Dublin, Ohio

TABLE 5. EXPERIMENTALLY DETERMINED CETANE NUMBERS OF RJ-6M WITH AND WITHOUT CETANE IMPROVER

	Cetane No.
RJ-6M (neat)	30
RJ-6M + 0.4 wt% amyl nitrate	39
RJ-6M + 0.8 wt% amyl nitrate	45

4. <u>Tetrahydronaphthalene (Tetralin*)</u>

1,2,3,4-tetrahydronaphthalene, commonly referred to as tetralin, is the product of the partial hydrogenation of naphthalene and is a solvent for oils, resins, asphalt, rubber, and waxes. The typical properties of tetralin solvent as supplied by the manufacturer are shown in Table 6. The Δ Hc and elemental analysis were determined at AFLRL.

Certain safety precautions are necessary with its use. On a one-time exposure, it may cause a mild irritation to the skin and eyes. Repeat

TABLE 6. TYPICAL PROPERTIES OF TETRALIN SOLVENT

Molecular Weight	132.2
Boiling Point (760 mm Hg), °C (°F)	207 (405)
Freezing Point, °C (°F)	-31 (-24)
Specific Gravity, 20°C	0.970
Vapor Pressure, mm Hg	
@ 100°C	27
@160 °C	221
Solubility in water	Negligible
Flash Point, tag closed cup, *C (*F)	82 (180)
Δ Hc Btu/1b (net)	17,578
∆Hc Btu/gal. (net)	140,295
Elemental Analysis, wt%	
Carbon	90.84
Hydrogen	9.16

*Du Pont Co., Wilmington, Delaware

exposure is more hazardous, as it can defat the skin, causing it to crack and lead to secondary infection. Studies with animals have caused liver and kidney damage and cataract formation by repeated oral or inhalation exposure. たいとう 古書言書を読

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5. Extract From Furfural Unit

It was noted in a recent report $(\underline{6})$ that the extract from the furfural unit in a lube plant has a relatively high heating value (154,000 Btu/gal.), but will probably vary, depending on the type of lube stock being processed. Several refineries in the United States use the furfural unit in their processes. Samples were requested from two of these refineries.

One sample (Nuflex⁽⁰⁾) has been received from its manufacturer* with a data sheet containing its typical properties. The Δ Hc was determined at AFLRL and found to be somewhat lower than expected. However, the extract would be expected to vary, depending on the lube oil being treated. The properties of the furfural extract are listed in Table 7. This product should be

 TABLE 7. PROPERTIES OF	FURFURAL EXTRACT	
API Gravity @ 15.6°C (60°F)	11.8	
Density, 1b/gal. @ 15.6°C (60°F)	8.223	
Viscosity @ 37.8°C (100°F), SUS	288	
Viscosity @ 98.9°C (210°F), SUS	44.4	
Aniline Point, °C (°F)	19.4 (67)	
Flash Point, COC, °C (°F)	207.2 (405)	
Refractive Index, 20°C (68°F)	1.5645	
Pour Point, °C (°F)	1.7 (35)	
Clay Gel Analysis		
Saturates, wt%	12.0	
Polar Compounds, wt%	8.6	
Aromatics, wt%	79.4	
Asphaltenes, wt%	0	
Elemental Analysis	Ŭ	
Sulfur, wt%	2.5-3.0	
Nitrogen, ppm	1500-2000	
Hydrogen, wt%	7.5-8.0	
Carbon, wt%	89-90	
Net Δ Hc	V7-7V	
Btu/lb (net)	17,540-17,590**	
Btu/gal. (net)	144,455-142,375	
•		

**Range given; actual value depends on percent of hydrogen present

*Ashland Chemical, Dublin, OH

relatively inexpensive, and further testing of the product is planned.

6. SRC-II (Solvent-Refined Coal II)

A process is under development for transforming high-sulfur coal into distillate fuels and various gaseous products. Plans have been made to construct a 6000-ton per day demonstration plant for DOE near Morgantown, WV, but as of this date these plans have been temporarily suspended. The use of SRC-II fuels for steam was expected to become competitive with petroleumbased fuels within 10 years. (7) A sample of a fuel oil blend (2.9:1 middle distillate:heavy distillate) was obtained from a solvent-refined coal pilot plant at Fort Lewis, WA. The analyses received with the sample were for a batch blend; not for the specific sample reported here. These analyses are listed in Table 8. Tests conducted at AFLKL indicated that the sample received was a representative one. 1

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TABLE 8. PROPERTIES OF SRC-II (2.9 to 1, Middle Distillate to Heavy Distillate)

Specific Gravity, 15.6°C (60°C)	0.999
*API at 15.6*C (60*C)	10.14
Viscosity	
@ 43.3°C (110°F), cSt	4.527
@ 98.9°C (210°F), cSt	1.289
Flash Point, ASTM D 93, °C (°F)	71.1 (160)
Heat Content	
Btu/lb (net)	16,540
Btu/gel.	137,895
Coal Tar Acid, m1/100 g	25.9
Sulfur, wt%	0.21
Ash, %	0.02
IBP, *C (*F), ASTM D 86	170 (338)
End Point, °C (°F)	371 (700)
Hydrogen*, wt%	8.64

*Determined at APLRL

The fuel listed in Table 8 was known to be highly aromatic and to have a very low cetane number. The cetane test engine would not run on SRC-II neat. Therefore, 3 wt% and 5 wt% amyl nitrate (a cetane improver) was added to samples of the above fuel oil blend, and cetane determinations were made at AFLRL. The results are shown in Table 9.

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		DETERMINATIONS CETANE IMPROVER	OF SRC-II AND BLENDED WITH DF-2
Wt% SRC (AL-8321-F)	DF-2 (AL-7225)	Wt% Amyl <u>Nitrate</u>	Resulting Cetane_Number
100.0			Engine would not run
	100.0		49
97.0		3.0	18
95.0		5.0	27
48.5	48.5	3.0	37
48.5	50.0	1.5	29
Indicate	s none of this	material used in	the blend.

Even with the amyl nitrate added, the cetane number of this SRC-II was still relatively low. A sample from this fuel (without cetane improver) was run in a CLR engine as a 50:50 mixture with diesel fuel on another program at Southwest Research Institute (SwRI). No difficulties were experienced on this brief run.

Another sample of SRC-II from the same source was almost entirely middle distillate. Cetane number was determined to be 16. The properties of this fuel, determined at AFLRL, which related to the high-energy fuel program are summarized in Table 10, and the complete inspection dats on this SRC-II middle distillate sample can be found in Appendix A, No. 1. The safety and toxicological aspects of solvent-refined fuels are given in Appendix A, No. 2.

TABLE 10. PROPERTIES OF SRC-II (Middle Distillate)

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Specific Gravity, 15.6/15.6°C (60°C)	0.98 40
Viscosity	
@ 37.8°C (100°F), cSt	3.83
@ 40.0°C (104°F), cSt	3.68
Flash Point, ASTM D 93, °C (°F)	80 (176)
Heat of Cowbustion	
Btu/1b	16,410
Btu/gal.	134,755
Hydrogen, wt%	8.64
Ash, wt%	0.0023
Cetane No.	16

7. Polycyclic Aromatic Blending Stock

Another refinery product that is relatively inexpensive and has a fairly high volumetric heat of combustion is a polycyclic aromatic blending stock. Its properties, as determined at AFLRL, are listed in Table 11.

TABLE 11. PROPERTIES OF POLYCYCLIC AROMATIC	BLENDING STOCK
Specific Gravity, 15.6/15.6°C (60°C) Heat of Combustion	1.0173
Btu/1b (net)	16,535
Btu/gal. (net)	140,340
Carbon, wt%	90.89
Hydrogen, wt%	7.70
FIA (Aromatics), wt%	97
Monoaromatics, wt%	7.2
Diaromatics, wt%	75.2
Triaromatics, wt%	1.4
Olefin, wt%	1.4
Saturates, wt%	1.5

8. Burner Oil No. 5

This fuel is included as a blending component because of its moderately high volumetric heat content and relatively low cost. Its properties, as determined at AFLRL, are listed in Table 12.

TABLE 12. PROPERTIES OF NO. 5 BUKNER OIL

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Cetane No.	41.0
Viscosity, 40°C (104°F), cSt	185.9
API Gravity, 15.6°C (60°F)	22.8
Density, g/ml, 15.6°C (60°F)	0.9170
Flash Point, ASTM D 93, °C (°F)	108 (227)
Sulfur, wt%	0.65
Pour Point, °C (°F)	21 (70)
Ash, wt%	0.248
Hydrogen, wt7*	13.86
Heat of Combustion	
Btu/lb (net)	17,995
Btu/gal. (net)	137,700
Copper Strip Corrosion, 50°C (122°F)	
ASTM D 130	la

*Calculated

9. No. 6 Fuel Oil (Bunker C)

This is a heavy residual oil that requires preheating before it can be burned. It is used as a fuel for industry and ships, and can be obtained from straight-run or cracked residiums in the refinery operation. This oil is comparatively inexpensive with a relatively high volumetric heat of combustion. The properties are listed in Table 13.

TABLE 13. PROPERTIES OF NO. 6 FUEL OIL

Cetane No.	26
Viscosity, 40°C (104°F), cSt	137,89
"API Gravity, 15.6°C (60°F)	13.2
Density, 15.6°C (60°F)	0,978
Sulfur, wtX	1.5
Pour Point, °C (°F)	-3 (27)
Ash, wt%	0.47
Hydrogen, wt ²	10.58
Heat of Combustion	
Btu/lb (net)	17,350
Btu/gal. (net)	141,575

10. <u>JP-9</u>

This is a three-component blend of JP-10 (65-70 wt%), RJ-5 (20-25 wt%), and methylcyclohexane (10-12 wt%). It has a net volumetric heating value of approximately 142,000 Btu/gal. with a flash point of 24° C. Only a limited amount of testing was done with this fuel.

B. Solid High-Energy Components

Some of the solid components described below and listed in Table 14 are materials that are soluble in other liquid components or in DF-2. They can be used to augment the energy content of the component or fuel in which they are dissolved. In cases where the solubility is not appreciable, slurries are a definite possibility, although one-phase liquid fuels are the more desirable. THE REAL PROPERTY.

1. Naphthalene

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<u>Soli</u>	d Component ^(a)	Density, °C	Net Heat of Combustion, <u>Btu/lb</u>
Na	phthalene	$1.178 \frac{15.56}{15.56}$ (b) (c)	16,720 (b)(c)
		1.162 $\frac{20}{4}$ (d)	16,725 (e)
Pt	enanthrene	1.179 25 (d)	15,935 (g)
		$1.063 \frac{15.56}{15.56} (f)$	
Ar	thracene	$1.283 \frac{25}{4}$ (h)(1)	l6,695 (g)
		$1.25 \frac{27}{4}$ (d) (j)	16,725 (k)
			16,770 (c)
SF	RC-1	1.2 (k)	15,215 (g)
	irbon Black (n) Raven 1170)	$1.80 \frac{15.56}{15.56}$ (e)	13,875 (g)
Note (a) (b)	Solid at room tempe	a - Petroleum Refining, 2nd	f Edition, 1970, American
(c)			
(d)		inth Edition, Merck & Co.,	
(e)		a in CRC Handbook of Chema West Palm Beach, FL.	latry & Physics, Jyth
(f)	-	cal Dictionary, Fifth Edit	tion, 1958, Reinhold Pub-
(g)		ermined at AFLRL.	
(h)	CRC Handbook of Che Beach, FL.	mistry & Physics, 59th Edd	ltion, CRC Press, West Palm
(1)		aic Compounds, Revised Edit	tion, 1953, Oxford Univer-
(ታ)	•	a in note (h) and correcte	ed to net value. Litera-
(k)		a in note (i) and correcte	ed to net value. Litera-
(1)		urement not given in refer	rence; from Table of Typi-

TABLE 14. SOLID HIGH-ENERGY COMPONENTS

 Temperature of measurement not given in reference; from Table of Typical Properties of SRC-I Product, Coal, Feed-Western Kentucky Bituminous, Synthetic Fuels Data Handbook, 2nd Edition, Cameron Engineers, Inc., Denver, CO.

- (m) Manufacturer (Cities Service Co., Columbian Div., Akron, OH) suggests this density for practical purposes in calculating formulations.
- (n) See Appendix H for the properties of the carbon ' cks used in this report.

Naphthalene can be derived from coal-tar oils or produced by dealkylation of a refinery stream containing heavy aromatics such as those found in a heavy reformate or from cycle stock from a catalytic cracker.

Naphthalene is used in the manufacturing of phthalic anhydride and 2-naphthol. It is the raw material used in the preparation of various pharmaceutical products and the making of synthetic organic dyes, and is also used as an intermediate in the preparation of tetralin and decalin. Its use as a moth repellant and insecticide has declined since chlorinated compounds were introduced for that purpose.

2. Anthracene



Anthracene is obtained from coal tar distillation. It is either insoluble or has very limited solubility in most organic liquids. The solubility of anthracene is discussed in Section III. D. la of this report.



Phenanthrene is an isomer of anthracene that is also obtained from coal tar. As discussed in Section III. D. 2a, it is more soluble than anthracene in all of the material tested at AFLRL.

4. <u>SRC-I</u>

The process for converting a high-sulfur, high-ash coal to a low-sulfur, low-ash solid fuel is known as the SRC-I process. The typical properties of SRC-I prepared from a western Kentucky bituminous coal are shown in Table 15.

Carbon, wt%	88.0
Hydrogen, wt%	5.9
Nitrogen, wt%	2.2
Sulfur, wt%	0.7
Oxygen, wt% by difference	3,1
Ash, wtX	0.2
Forms of Sulfur	
Sulfate, wt%	
Pyritic, wt%	
Organic, wt%	0.7
Heating Value, Btu/1b	16,250
Pusion Point, "F	
(gradient bar)	350
Density, gm/cm	1.2

TABLE 15. TYPICAL PROPERTIES OF SRC-I (7)

It is assumed that the heating value listed in Table 15 is the gross heating value. The gross Δ Hc was determined by running a sample of SRC-I at AFLRL. The sample had a gross Δ Hc of 15,750 Btu/lb and a net value of 15,215 Btu/lb.

5. Carbon Black

The use of carbon black in preparing slurries with various liquid hydrocarbons to increase the vol . energy content is still a good possibility. The properties of carbon blacks and the slurry technology have been described in Reference 3.

C. Blending and Characterization of Liquid Fuels

The properties of the blended fuels prepared from the various components were compared to a DF-2 Diesel Control Fuel* that meets the Environmental Protection Agency (EPA) specification for emission testing**. The properties of this baseline fuel are listed in Table 16, while the properties of

* From Phillips Chemical Company, Bartlesville, OK **As described in Chapter One, EPA, Part 86.177.6

Gravity, [•] API	34.8
Specific Gravity	0.8509
Distillation, D 86 °C (°F)	
IBP	196 (384)
5	212 (414)
10	222 (432)
20	234 (454)
50	266 (510)
90	301 (574)
95	309 (588)
EP	320 (608)
Recovery, X	98.5
Viscosity, cSt @ 40°C	2.48
Flash Point, °C (°F)	71.7 (161)
Flash Point, °C (°F) Pour Point, °C (°F)	-21.7 (-7)
Hydrocarbon Type (FIA), vol%	
Aromatics	29.6
Olef ins	1.6
Saturates	68.8
Elemental Analysis, wt%	
Carbon	86.63
Hydrogen	12.85
Sulfur	0.34
H:C Atom Ratio	1.78
Heat of Combustion	
Gross, Btu/lb	19,265
mg/kg	44.81
Net, Btu/1b	18,095
mg/kg	42.08
Btu/gal.	128,465
Accelerated Stability, mg/100 ml	0.3
Existent Gum, mg/200 ml	3.9
Cetane No.	48.0

TABLE 16. LABORATORY INSPECTION OF BASELINE FUEL* (CONTROL FUEL)

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other diesel fuels used in this report can be found in Appendix B. The volumetric heat content of a fuel serves as a guide to forecast the fuel's ability to increase the range of a vehicle in which it is used, and should be a useful tool in deciding which fuels should advance to the next screening stage, i.e., engine testing.

The composition of the fuel blends in this report are listed as weight percent. In the blending of fuels, it is sometimes more convenient to measure the fuel components by volume and then weigh them. For this reason, the weight percents usually result in fractions of whole numbers.

Since a high-energy fuel candidate should have a high volumetric heat of combustion, and heats of combustion are usually expressed in Btu's per pound, it would be convenient to have a conversion chart that would indicate quickly if a fuel might qualify as a high-energy fuel (knowing the density and Btu's/lb). Figures 2 and 3 can be used for this purpose. These figures illustrate the large effect that slight changes in density make upon the volumetric heat content of the fuel.

1. SRC-II

SRC-II, middle distillate, has a volumetric \triangle Hc of approximately 134,750 Btu/gal. Since the goal of the high-energy fuel program is a 10-percent increase in range without any increase in fuel tank volume, SRC-II needs to be augmented with a component of higher volumetric heat content such as RJ-5.

a. Augmenting SRC-II With RJ-5

Table 17 shows the experimental results of augmenting SRC-II fuel with RJ-5. Since both components have low cetane ratings, it would probably be necessary to add a cetane improver, such as amyl nitrate, when using it in a diesel engine. The disadvantage of using RJ-5 is its high price.

b. Augmenting SRC-II With JP-10

Blending SRC-II (middle distillate) with JP-10 had only a slight effect on increasing its energy content (volumetric heat of combustion). For example, an equal mixture by weight of JP-10 and SRC-II, with 1 percent amyl nitrate added, increased the energy content by only 0.3 percent. The low heat content of amyl mitrate (10,600 Btu/1b) contributed to hindering the increase in energy. This cursory study is included in Table 18.





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FIGURE 3. HEAT OF COMBUSTION OF FUELS-APPROXIMATE BTU-GRAVITY RELATION (FROM "TABLES OF USEFUL INFORMATION," EXXON CORP., LUBETEXT D 400, JANUARY 3, 1973)

Heat Content, X Improvement Over Control Fuel(b)	4.9	6.7	8.5	10.3	8.1	10.5	14.1	19.1	24.9
Heat of Combustion (net) <u>Etu/1b</u> <u>Btu/gal.</u> 0	134,755	137,090	139,430	141,660	138,830	141,955(c)	146,585	153,050	160, 390
Reat of Comb Etu/1b	16.412	16,540	16,595	16,835	16,484	1	17,025	17,330	17,735
Viscosity @ 40°C, cSt	1	3.97	ł	ł	4.03	1	5.26	ł	l
Bydrogen, Nt I	8.64	8.75(c)	8.08(c)	8.93(c)	9.16	1	9.16(c)	9.45(c)	9.74
Deumsity, 15.56°C (60°P) E/mil	0.9840	¥696.0	1.0585	1.0084	1.0093	1	1.0319	1.0585	1.0820
Amyl Nitrate, VtZ	1	ł	1.00	1	2.98	2.94	1.00	1.00	
2-2	1	10.89	24.50	26.82	25.24	35.64	49.50	74.50	100.00
SRC-II (a) . BLT	100.0	11.68	74.50	73.18	71.78	61.42	49.50	24.50	

TABLE 17. EXPERIMENTAL RESULTS SHOWING THE EFFECT OF AUGMENTING SRC-11 MIDDLE DISTILLATE FUEL WITH RJ-5

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AFTRL Fuel No. 8321F. Control Puel, ABC (net) 128,465 Btu/gal. Calculated from the individual components. **3**23

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Cetane No.(a) (c) (c) (c) 30 30 45 (c) (c) (c) 30 (c)
<pre>% Improvement Volumetric ARC Volumetric ARC 6.0 6.0 4.4 11.7 -31.1 16.8 16.8 16.6 16.5 10.8 7.8</pre>
A Bc (net) Btu/1b B.u/gal. 17,030 136,215 16,545 134,080 16,545 134,080 18,290 143,450 17,970 150,070 17,970 150,070 17,911(b) 149,605(b) 17,911(b) 149,605(b) 17,911(b) 149,555(b) 16,801(b) 138,555(b)
Anyl Mitrate, Witz 1.0 1.0 0.4 0.8
19-10. 19.5 24.5 24.5
49.5 49.5 74.5 50.0
RI-64, HCZ 130 99.6 25.0

SRC-IL AUCHENTED WITH JP-10 AND RJ-6M TABLE 18.

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Determined at AFLRL. Calculated from the individual components. Not determined.

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c. Augmenting SRC-II With RJ-6M

In Section III.A.3, RJ-6M was discussed. Only a limited amount of work has been done with this fuel since it was received just prior to the writing of this report. It is much less costly than RJ-5 and has a rather high heat content of 150,070 Btu/gal. (16.8 percent improvement compared to control fuel).

Some calculated values of blends of RJ-6M with SRC-II are listed in Table 18. For example, a 50:50 mixture by weight results in a blend having a 10.8 percent increase in volumetric heat content when compared to the control fuel. The cetane number on RJ-6M was determined at AFLRL and found to be 30. Addition of 0.4 and 0.8 wt% amyl nitrate increased the cetane number to 39 and 45, respectively.

RJ-6M <u>must</u> be considered as an excellent candidate fuel at this point in its testing.

2. Tetralin

a. Blands With DF-2

Tetralin has a low cetane number and would require either a substantial amount of cetane improver or would need to be blended with another component in order for it to be run in a dissel engine. Blends of tetralin with DF-2 are shown in Table 19. The blend containing 75 wt% tetralin, 24 wt% DF-2 (Cet 1-H; 53 cetane number) and 1.0 wt% cetane improver was run in the Petter engine without difficulty.

b. Blends With No. 6 Burner Fuel

A 50:50 blend by weight of tetralin and No. 6 burner fuel would probably need a cetane improver, but indications are that it would have a 10-percent improvement in heat content over that of the control fuel.

$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	No. 6 rner Fuel, WtZ	No. 6 Burner Fuel, Tetralin, VtZ VtZ	DF-2(a) WtZ	Amyl Nitrate VLZ	Density, 15.56°C (60°F) g/ml	Hydrogen, UtZ	Heat of Combustion (net) Btu/1b Btu/gal.	istion (net) Btu/gal.	<pre>% Improvement Over Control Fuel(b)</pre>
97.0 3.0 0.9709(b) 17,370 140,990(b) 75.0 24.0 1.0 0.9386 10.11 17,815 139,510 80.0 19.0 1.0 0.9491 9.92 17,735(b) 140,440 0.9779 10.58 17,735(b) 141,575 9.0 1.0 0.9642 10.61 17,559 141,575 50.0 0.9740(b) 17,465(b) 141,945(b)		100.0		ł	0.970	9.16	17,580	142,275	11.7
75.0 24.0 1.0 0.9366 10.11 17,815 139,510 80.0 19.0 1.0 0.9491 9.92 17,735(b) 140,440 0.9779 10.58 17,350 141,575 9.0 1.0 0.9642 10.61 17,559 141,275 50.0 0.9740(b) 0.9740(b) 17,465(b) 141,945(b)		97.0	1	3.0	0.9709(b)	ł	17,370	140,990(b)	9.5
80.0 19.0 1.0 0.9491 9.92 17,735(b) 140,440 0.9779 10.58 17,350 141,575 9.0 1.0 0.9642 10.61 17,559 141,275 50.0 0.9740(b) 17,465(b) 141,945(b)		75.0	24.0	1.0	0.9386	10.11	17,815	139,510	8.6
0.9779 10.58 17,350 141,575 9.0 1.0 0.9642 10.61 17,559 141,275 50.0 0.9740(b) 0.945(b) 117,465(b) 141,945(b)		80.0	19.0	1.0	0.9491	9.92	17 , 735(b)	140,440	9.3
9.0 1.0 0.9642 10.61 17.559 141.275 50.0 0.9740(b) 17.465(b) 141.945(b)	100.0		1	ļ	0.9779	10.58	17,350	141,575	10.2
50.0 0.9740(b) 17,465(b) 141,945(b)	0-06	1	0.6	1.0	0.9642	10.61	17,559	141,275	10.0
	50.0	50.0	1	1	0.9740(b)	8	17 , 465(b)	141,945(b)	10.5

TABLE 19. LIQUID FUEL BLENDS CONTAINING NO. 6 BURNER FUEL, TETRALIN, AND DF-2

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Contraction of the
3. No. 6 Burner Fuel

a. Blend of No. 6 Burner Fuel With Cat 1-H Diesel

Another possible high-energy fuel is a blend of No. 6 burner fuel with another component that would make the properties of the burner fuel acceptable. For example, a blend containing approximately 90 wt% No. 6 burner fuel, 9.0 wt% DF-2 (Cat 1-W) and 1.0 wt% amyl nitrate resulted in a fuel that had a net volumetric heat of combustion of 141,275, or a 10 percent improvement over the control fuel. The viscosity as measured by ASTM D 445 decreased from 185.9 cSt at 40°C ($104^{\circ}F$) in the No. 6 burner fuel to 71.5 cSt at 40°C in the blend. The cetane number of the No. 6 burner fuel was 26, while that of the Cat 1-H was 53, which means the blend would be upgraded in cetane number even without the cetane improver (see Table 19). The properties of the Cat 1-H fuel are in Appendix B, No. 1.

4. RJ-6M

Since this is one of the most recent candidate fuels, blending studies with other fuel components have not yet been started. However, from the limited data available and from the Petter engine tests (to be described later), this synthetic hydrocarbon gives all indications of being an excellent high-energy fuel and/or component.

5. Furfural Extract

This material is also a recent candidate for the high-energy fuel program and its testing has been limited thus far.

D. Fuel Preparation and Characterization With Solid High-Energy Components

1. Insoluble Components Used in Slurries

a. <u>Anthracene</u>

Anthracene is a good candidate for a high-energy component since it has a

relatively high density (1.283 g/ml at 25°C) and experimental measured net Δ Hc of 16,695 Btu/lb. This experimental value is in fair agreement with the literature value measured at 25°C of 16,770 Btu/lb. Unfortunately, it is not very soluble in most common hydrocarbons.

A brief examination of the chemical literature and cursory solubility studies in the laboratory indicated that anthracene had no appreciable solubility in the various organic hydrocarbons in which it was tested. The laboratory studies included tetralin, DF-2, JP-10, RJ-5, mixed naphthalenes, FRF, and SRC-II.

In the literature examined, no appreciable solubility was found in methyl naphthalene, m-cresol, nitrotoluene, decahydronaphthalene, 1 methyl-2 pyrrolidinone, 1,4 dioxane, dibutyl cellosolve. In nitrobenzene, the solubility is 1.86 g/100 ml. Merck Index (9th Edition) reports 1.0 gram anthracene to be soluble in 67 ml absolute alcohol, in 70 ml methanol, 62 ml benzene, 85 ml chloroform, 200 ml ether, 31 ml carbon disulfide, 86 ml carbon tetrachloride, and 125 ml toluene.

Because of the potential energy content, slurries of anthracene in DF-2, tetralin, and RJ-5 were prepared. Each consisted of 49.85 wt% anthracene and 49.85 wt% of the fuel and 0.3 wt% aluminum octoate to prevent settling. Table 20 shows the resulting experimental heat of combustion of these

	TA	BLE 20. ANTHRACI	ENE SLURRIES ^(a)	
		Density 15.6/15.6 g/ml (Experimental)	Net ∆Hc Btu/gal. (Experimental)	X Improvement Over ΔHc in Net Control
Anthracene:	DF-2	0.9999	147,415	14.7
Anthracene:	Tetralin	1.0583	149,870	16.7
Anthracene:	,RJ-5	1.1501	165,110	28.5
Control DF-2	(0)	0.9874	128,465	

 (a) All slurries have 0.30% aluminum octoate added as suspension agent and are equal wt% (49.85) of anthracene and the other component.
 (b) AFLRL Fuel No. AL-9979-F (Control Fuel). slurried fuels and the percent improvement in the volumetric heating value over a typical DF-2. Because all the slurries in Table 20 show improvement over the "control diesel fuel," slurries of this type should be worthy of further study and testing.

b. Carbon Black

Experience with slurry fuels at AFLRL and other research being conducted at SwRI (a) have indicated several problems that need to be addressed before these fuels can be fully utilized. Difficulties are usually encountered when the carbon concentration in slurries is greater than 10 wt%, and the problem becomes correspondingly more troublesome as the carbon black concentration is increased. The problems alluded to are incomplete combustion, decreased thermal efficiency, degraded atomization, high viscosity, settling, and operational difficulties, such as plugging of the injector nozzle or sticking of the nozzle valve. However, great advances have been made in carbon slurry technology during the past few years, especially by researchers interested in coal slurries and by those working in government-sponsored research on the cruise missile.

Two recent important achievements (see 1 and 2 below) have been made recently to help promote carbon slurried fuels.

(1) High carbon loading at relatively low viscosity

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A style c \sim fuel containing 48 percent carbon black (SL-90) in JP-10 (with 3 it a surface \sim was received from Ashland Chemical Co. Its viscosity at ambi $\sim - \pi \omega_{PE}$ for the is such that this slurry flows easily. The data furnish with \sim mple are shown in Table 21. The high volumetric heat of combustion (168,185 Btu/gal.), approximately 31 percent higher heating value than the control DF-2, should make it worthy of consideration as a highenergy fuel candidate. Photomicrographs of the fuel are shown in Figure 4.

⁽a) U.S. Department of Energy, Office of Transportation Programs, Contract No. DE-AC04-79CS54240-001.

TABLE 21. PROPERTIES OF SL-90 CARBON SLURRY FUEL

Nominal wt% Carbon Black	48
Nominal wt% JP-10	49
Nominal wt% Surfactant	3
Analyzed wt% Carbon Black	47.2
Density @ 25°C, lb/gal.	10.09
Net Heat of Combustion, Btu/lb	16,371
Net Heat of Combustion, Btu/gal.	165,183
Flash Point, °C (°F)	53.3 (128)

Contraves RM-30 Viscosity:

<u>Temp, °C</u>	Shear Rate, Sec ⁻¹	<u></u>
25	1	663
25	10	182
25	100	83
25	600	53
-20	1	1562
-20	10	478
-20	100	241
-20	600	285

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b. 430X MAGNIFICATION

FIGURE 4. PHOTOMICROGRAPHS OF 48 WEIGHT PERCENT CARBON BLACK SLURRY IN JP-10

A representative of the manufacturer indicates that a carbon slurry of a similar concentration in DF-2 could easily be prepared using this proprietary method of preparation. If so, this should be a good candidate fuel available at a relatively low cost. 「「「「「」」」」

(2) "Catalyzed" carbon black

Previous engine studies have indicated a problem with achieving full utilization of the carbon when running in the CLR engine. Use of a catalyst could rossibly aid in its combustion and thus increase carbon utilization. The transition metals iron, nickel, and manganese were originally selected as catalyst candidates. At the recent Workshop on the Combustion of Slurry Fuels at the 17th JANNAF Combustion Meeting, it was stated that lead had been successfully used in the catalysis of carbon particle combustion in a turbine combustion chamber. Therefore, lead (as lead acetate) was added to the list of catalyst candidates. Deposition of the metals on the carbon black was achieved by dissolving the metallic compound in water [in methanol for iron (ferrocene)], adding the carbon black, and stirring for at least 1 hour. The carbon black was removed by filtration and then dried. Since the carbon black was extremely absorbent and the compounds containing the potential catalyst were highly polar, the deposition occurred without difficulty.

If 100 percent of the metal were deposited, the concentration of the carbon black would be 0.30 wt% for the iron, nickel, and lead, and 0.36 wt% for the manganese. The amount deposited has not yet been quantitatively determined, but X-ray fluorescence indicates the amount is relatively large. Washing of the carbon blacks with diesel fuel did not remove the deposits.

The above-described successful deposition work was first done on a relatively small scale (40 grams per test). Afterward, four larger batches (approximately 750 grams each), each containing a different metal, were processed. With this "catalyzed" carbon, four test fuels (8 liters each) were prepared containing 10 wt% carbon black in diesel fuel.

Engine tests with these fuels are described later.

(3) <u>Settling studies</u>

The freezing tube technique for measuring settling in carbon slurries was described in the first interim report on this program and is repeated in Appendix C of this report. The only difficulty encountered with this method thus far is that it tends to give slightly high values for the recovered carbon. These values appear to be high but correct relative to each other. This could be caused by the incomplete removal of the last traces of diesel fuel absorbed by the carbon black. A brief study was conducted to see if this rationale might be correct.

(a) <u>Refinement of the freezing tube technique</u>

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A fuel was prepared in the attritor to contain 20.0 wt% carbon black, 1.0 wt% lecithin, and 79.0 wt% diesel fuel. The resulting slurry was placed in a settling tube for 48 hours at ambient temperature. The tube was then frozen and sliced into eight equal sections, and the percent carbon was determined, using the method mentioned above. The carbon recovered from the top four sections contained 20.57, 20.55, 20.58, and 20.58 wt% carbon, respectively. The average is 2.85 percent higher than the 20.0 wt% that was used in the preparation. The carbon samples recovered from the four sections were then placed in a Soxhlet extraction thimble that had been prewashed and dried to constant weight with methylene chloride. The carbon (17.3760 g) in the thimble was then continuously extracted with methylene chloride for eight hours. The carbon was dried and reweighed (16.7320 g) with a loss of 0.6440 g noted. This loss represents 3.7 percent of the total at the start of the extraction. Upon evaporation of the methylene chloride, 0.3888 g of lesidue was recovered, or 2.2 percent of the total at the start. The 2.2 percent of material recovered from this carbon compared favorably with the 2.85 percent high value that was observed when the carbon was originally determined by the freezing tube method. The residue recovered from the methylene chloride was essentially DF-2.

The four lower sections contained 20.60, 20.60, 20.57, and 20.62 wt% carbon, respectively. The average of these four sections was 20.60 wt% carbon, or

3.0 percent higher than the 20.0 wt% used to make the preparation. The same extracting procedure used with methylene chloride was then used with pentane. A 3.0-percent loss in weight was noted, and a residue was recovered upon evaporation that represented 1.1 percent of the total carbon at the beginning of the extraction. This residue was also found to be DF-2.

It would appear that methylene chloride is more effective in removing the diesel fuel from the carbon and will be used in place of pentane in future studies using the freezing tube technique.

(b) Effect of temperature on the stability of carbon slurries

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Prior to the refinement in the freezing tube technique for measuring settling in carbon slurries, the effect of temperature on the stability of these slurries was examined.

A slurry containing 20 percent carbon black, 79 percent DF-2, and 1 percent lecithin was prepared in the attritor. Settling data were collected on a sample of the above fuel that had stood in a 100°C oil bath for 48 hours and a similar sample standing at room temperature (22°C) for the same period of time. The samples were analyzed by the freezing-tube technique.

It would appear from Figure 5 that about 4 percent more carbon had settled out of the top section of the sample at 100°C compared to the room temperature sample. This decrease of carbon in the top section of the 100°C sample is reflected in the increase of the carbon content of the lower sections.

(c) Stability of FRF-carbon slurries

Carbon black was added to a fire-resistant diesel fuel (FRF) so that the resultant fuel contained 20 wt% carbon black, 67.2 wt% DF-2, and 4.8 wt% water. Settling data were collected on samples after approximately 1, 15, and 38 days standing at ambient temperature. The samples were analyzed by the freezing tube technique prior to its refinement. As mentioned above.



FIGURE 5. EFFECT OF TEMPERATURE ON THE STABILITY OF SLURRIED FUELS (20% Carbon, 1% Lecithin in DF-2, 48 Hour Settling Period)

the original procedure tends to give slightly high values for the recovered carbon. Although these values appear to be somewhat high, they are correct relative to each other. The high value is attributed to the incomplete removal of the last traces of diesel fuel that is absorbed by the carbon black. The refinement to this technique mentioned previously eliminates this problem. Even without the refinement, it is an excellent indicator of stability since all samples are treated in an identical manner.

The data from the FRF/carbon black stability study are plotted and shown in Figure 6. The mixture is fairly stable for at least 1 day as indicated from the carbon content in the six sections. For example, the carbon percentages found after 1 day, from top to bottom, were 22.0, 22.8, 23.1, 23.2, 23.7, and 25.5. It would appear that only slight settling has occurred.

c. SRC-I

A cursory search was made for a hydrocarbon in which SRC-I would have an



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FIGURE 6. SETTLING DATA: 20 PERCENT CARBON BLACK IN FRF

appreciable solubility. No such hydrocarbon was found at that time but the search is continuing.

A slurry containing 20 wt% SRC-I in DF-2 was prepared, and the gross experimental heating value was 18,860 Btu/lb. Using the literature value for the density of SRC-I and the experimental density of DF-2, and assuming the densities are additive, the net volumetric heating value for the slurry was found to be 136,865 Btu/gal., a 6.5-percent improvement over DF-2.

2. Solid High-Energy Components That Have Appreciable Solubility

a. Solubility Measurements of Naphthalene and Phenanthrene

Naphthalene and phenanthrene are two solid (at room temperatures) hydrocarbons that have potential as high-energy fuel components. Phenanthrene is an isomer of anthracene but has the distinct advantage in that it is not as insoluble in organic hydrocarbons as anthracene. For example, in tetralin, anthracene was found to have very little solubility, while the solubility of

phenanthrene was experimentally found to be approximately 43.6 grams of phenanthrene per 100 grams of tetralin at 21°C (Table 22). Therefore, a sample containing 30.37 wt% phenanthrene in tetralin (approximately the maximum concentration) was prepared. The gross experimental heat of combustion was 18,135 Btu/1b. Using the experimentally determined density of 0.9930 g/ml at 36°C corrected to 1.006 g/ml at 15.6°C, the net volumetric heating value was calculated to be 146,025 Btu/gal. This would represent a 13.7-percent improvement over the typical diesel fuel. In order for such a mixture to be run in a diesel engine, cetane improver would be necessary. Phenanthrene contribution to the volumetric heat content of a blend or slurry would be less than that of anthracene.

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	Phenenthrene, g/100 g	Naphthalene, g/100 g
Tetrahydronaphthalene (tetralin)	50.0	43.6
DF-2 ^(a)	24.1	9.9
JP-10	29.5	12.0
RJ-5	(c)	15.3
Mixed Naphthalenes ^(b)	29.7	27.4
SRC-II (middle distillate)	29.3	29.2
FRF	15.3	8.2

TABLE 22. SOLUBILITIES^(a) OF NAPHTHALENE AND PHENANTHRENE IN VARIOUS ORGANIC HYDROCARBONS AND RELATED COMPOUNDS AT 20°C (68°F)

(a) AFLRL Fuel No. AL-9979-F (Control Fuel).

(b) AFLRL Fuel No. AL-9509-A.

(c) Not determined.

The solubility of phenanthrene in DF-2 (control fuel) was experimentally found to be 9.9 g of phenanthrene per 100 grams of DF-2 at 20°C. A sample was prepared containing 9.0 g phenanthrene in 91.0 g of DF-2 and the experimental Δ Hc was determined. Using the experimentally determined density of

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0.8693 g/ml at 15.6°C, the net volumetric heating value was found to be 133,160 Btu/gal., an improvement of 3.6 percent over neat diesel fuel. Probably no cetane improver would be required with this fuel. It would be necessary, however, to avoid temperatures much below 20°C, because the phenanthrene would crystallize and settle out.

The solubility of naphthalene and phenanthrene in various organic hydrocarbons and related compounds are shown in Table 22.

b. <u>Blending of Solid Components Into Liquid Fuels</u>

Various fuels have been prepared using these two high-energy components. Table 23 compares these fuels and their AFLRL-determined heats of combustion with the control DF-2. The experimentally determined cetane numbers of fuel blends containing naphthalene and phenanthrene are shown in Table 24.

E. Impact Dispersion Tests to Study Fire-Safety Characteristics of Energy-Augmented Fuels

1. RJ-5

Several high-energy or related fuels were examined for their fire-safety characteristics by the AFLRL Impact Dispersion Test. The results from these tests indicated that RJ-5 and some of its blends had some good fire-hindering properties.

Impact dispersion tests were conducted on RJ-5 and some of its blends.

- (a) RJ-5 (neat)
- (b) RJ-5: DF-2 Blend (50 vol%; 50 vol%)
- (c) RJ-5: FRF Blend (30 vol%: 70 vol%)*

^{* =} For a description of these tests and the rating system, see Appendix D.
** = 56.05 wt% RJ-5; 43.95 wt% DF-2 (AFLRL Fuel No. AL-8821-F)

 ^{= 34.65} wt% RJ-5 and 65.35 wt% FRF (FRF prepared using DF-2 with AFLRL Fuel No. AL-8821-F). The resulting net volumetric ΔHc was 128,570 Btu/gal.; about equal to control fuel (128,465).

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TABLE 23. COMPARISON OF THE VOLUMETRIC HEATS OF COMBUSTION OF VA	LIQUID FUELS CONTAINING NAPHTHALENE
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TABLE 2	IND
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Percent Increase in Volumetric ABC Over Control DF-2	10.2	1.11	0.61	1.11	13.7	1.6	3.2	8.8	-6.8	8.5	(P) 6.8	21.7	10.7	4.11	13.1	-3.8	-5.3	14.6	9.7
mtel utton Btu/gal.	141.555	142,715	145,190	142,755	146,025	091,001	132,545	139,745	119,765	139,400	212,761	164.070	142,200	143,065	145, 140	123,625	121.665	147,185	140,920
Experimental ABC Combustion Btu/1b Btu/	17,810	18,695	16, 790	16,460	295,11	18,335	18,155	17.525	16,135	16,545	18, 375 ^(b)	17,996	17,275	17,960	16,800	16,985	16,420	16,985	16,665
hydro ge a, vc2	10.94	10.66	7.01	11.7	8, 10 ^(b)	12. 20 ^(b)	11.58 ^(b)	10.63	11.92	8.052 ^(b)		9.20 ^(b)	8.20	10.58	7, 39 ^(b)	11.78	12.26	6.96	8.17
Denatry, 15.56°C (60°F), g/al	0.9524	0.9651	1.0363	1,0394	1.006	0.8693	0.8750	0.9556	0.8894	1.0097	0.8.35 ^(b)	1.0926	0.9864	0.9546	1.0367	0.8768	0.8881	1.0365	1.0134 ^(b)
Liquid Components. vtZ	80.0	77.6 3.0	80.0	77.6 3.0	69.63	90.99	80.58	72.0 3.0	85.6	80.27	85.00	86.70	66.67	11.22	78.00	86.70	92.50	78. 50	77.40
Liquid	.	4 A	ડ	4 ئ	7	P	.	48	2	3	Ŧ	1	7	ł	ያ	٤	٤	ŝ	\$
Phenanthrene, vcz	•	19.4	20.02	19.4	16.00	10.6				19.73		6.61			22.0		7.5	21.5	22.6
Maphchalene. vcX	20.0						19.42	25.0	14.4		15.0		11.11	22.78		13.30			

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See Code Letter 3

- Mayl Mitrate (cetame improver) krommiic Blending Stock fetralin 2-10 1101110
- DE-2 (control fuel) AFLML Fuel No. AL-9979-F
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- XMC-II (∎1441e distillate) XF-2 (Catl-H) AFLAL fuel No. AL-10115-F 11
 - 2
- Calculated from individual components 2

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TABLE 24.	CETANE	NUMBER	s op	FUEL	BLENDS	
CONTAINING	G NAPHTI	IALENE	or pi	HENANT	THRENE	

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Fuel Compositi	lon wtX	AFLRL Experimental <u>Cetane No.</u>
DF-2 ^(a)	91.0	
Phenanthrene	9.0	45
DF-2 ^(a)	80.6	
Naphthalene	19.4	43
Tetralin	65.3	
Naphthalene	32.7	
Amyl Nitrate	2.0	24
Tetralin	68.3	
Phenanthrene	29.7	
Amyl Nitrate	2.0	24

Examination of the video record of these tests showed no ground fires and only a medium-size fireball which is equal to a "D" rating. This is the rating that all the FRF fuels (84 vol% DF-2; 10 vol% H_2O ; 6 vol% surfactant) have received. The fact that these fuels containing RJ-5 did not sustain pool burning could reflect the effect of the high flash point of RJ-5 (111°C), since the fuels were dropped at 77°C onto the impact plate at 93.3°C.

2. <u>DF-2</u>

It is noted in Table 25 that the amount of RJ-5 needed to eliminate pool burning of DF-2 (AL-8821-F) in this test has been found to be between 35 to 50 volume percent. The blend containing 30 vol% RJ-5 and 70 vol% DF-2 has a coalesced fireball with simultaneous light pool burning immediately after impact, thus an "E" rating. The intensity of this ground fire then increased with time. DF-2 (neat) had an "E" rating.

TABLE 25. IMPACT DISPERSION TEST RESULTS AND FUEL PROPERTIES (a)

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Fuel	Impact Dispersion Test_Rating(b)		Density (15.56°C), g/ml	Viscosity 40°C, CSt
DF-2 (neat) (c)	E	130,620	0.8484	3,20
SRC-II (neat) (d)	E	134,760	0.9840	3.68
JP-9 (neat)	E	143,445 (e)		2.49
JP-10 (neat)	E		••••	
JP-10 microemulsion (f)	E	126,190	0.9533	(g)
RJ-5 (neat)	D	160,390	1,0837	13.27
50 vol% RJ-5; 50 vol% DF-2				
(56.05 wt% RJ-5; 43.95 wt% DF-2)	D	144,220	0.9657	5.78
30 vol% RJ-5; 70 vol% DF-2	-			5470
(35.38 wt% RJ-5; 64.62 wt% DF-2)	E	139,550(h)	0.9316(h)	(0)
35 vol% RJ-5; 65 vol% DF-2	-		01/210(11)	(8)
(40.75 wt% RJ-5; 59.25 wt% DF-2)	D	141,140	0.9328	4.73
25 vol% RJ-5; 25 vol% SRC-II	D	141,140	0.7520	4.73
(76.77 wt% RJ-5; 73.14 wt% SRC-II)	D	141,630	1.0084	4.40
75 vol% RJ-5; 25 vol% SRC-II	D	141,030	1.0004	4.40
(76.77 wt% RJ-5; 23.23 wt% SRC-II)	D	156,080	1.0632	8.14
30 vol% RJ-5; 70 vol% FRF		190,000	1.0052	0.14
(34.65 wt% RJ-5; 65.35 wt% FRF)	D	128,565	0.9382	(-)
(34.65 W(% KJ=5; 65.55 W(% FRF)	D(1)	116,825(h)	0.9382 0.8747(h)	(g)
	D(1)	110,025(1)	U.8/4/(n)	(g)
46 vol% FRF; 54 vol% SRC-II	P	196 216(6)	0 0330/6)	(-)
(43.08 wt% FRF; 56.91 wt% SRC-II)	E	126,315(h)	0.9338(h)	(g)
46 vol% FRF; 54 vol% JP-10		100 100 (1)	• • • • • • • •	
(44.24 wt% FRF; 55.76 wt% JP-10)	E	128,135(h)	0.9095(h)	(g)
 (a) Experimentally determined at (b) Impact Dispersion Test Rating A = No pilot flame enlargemen B = Pilot flame dimentions le C = Pilot flame dimensions mode D = Pilot flame totally obscurate E = Coalesced fireball with an endation 	at ess than doubled ore than doubled ored by transies	i. i. nt mist fireb		
Test Conditions: Sample 76.7 Impact Plat	°C (170°F) e 93.3°C (200°)	?)		
(c) AFLRL Fuel No. AL-8821-F, Fla with this table where DF-2 is properties.				
(d) Middle Distillate, AFLRL Fuel used in all blends in this ta				.76°F)
(e) Calculated from AFLRL data.				
(f) 83.5 wt% JP-10; 10 wt% H_2 0; 6	.0 wt% aurfact	ant: 0.5 wt7	cetane impr	over.
(g) Not determined.				
(h) Calculated from components.				
 (1) FRF made with DF-2 having AFL based on other FRF tests with 			. Rating g	iven

The properties of DF-2, RJ-5, and their equal blends are listed in Table 26. It can be seen from this table that a 50:50 volumetric blend of DF-2 and RJ-5 results in a fuel that shows an improvement in the volumetric heat of combustion of 12.3 percent when compared to the control DF-2. 第三 いう

	$DF-2^{(a)}$	RJ-5	50 Vol% RJ-5; 50 Vol% DF-2(b)
Btu/gal. (net)	130,615	160,390	144,220 ^(c)
Btu/lb (net)	18,450	17,740	17,900
Density, 15.6°C (60°F)	0.8484	1.0837	0.9657
Viscosity, cSt, 40°C	3.2	13.27	5.78
Flash Point, °C (°F)	71.7 (161)	111 (232)	86.1 (187)
Freeze Point, °C (°F)		-53.9 (-65) ^(d)	2 (36)
Pour Point, °C (°F)	-10 (-14)		

TABLE 26. PROPERTIES OF DF-2, RJ-5, AND THEIR EQUAL BLENDS

(a) AFLRL Fuel No. 8321-F.--see Appendix B, No. 2 for its properties.

(b) 56.05 wt% RJ-5 and 43.95 wt% DF-2.

(c) This represents a 12.3 percent increase over that of control fuel.

(d) Literature value (exp. value at another lab).

3. SRC-II

The blends of SRC-II and RJ-5 at SRC-II concentrations of 25 and 75 vol%, respectively, received "D" ratings. The data in Table 25 indicate that 25 vol% RJ-5 will eliminate pool burning of SRC-II. Lesser amounts of RJ-5 with SRC-II have not yet been tested. SRC-II (neat) received "E" ratings.

4. Other Fuels [JP-10 (neat), JP-9, JP-10/H O Microemulsion]

Other fuels to receive "E" ratings were JP-10 (neat), JP-9 (neat), and JP-10/ H_0 0 microemulsion (see Table 25).

JP-9 is a three-component blend containing JP-10 (65-70 wt%), RJ-5 (20-25 wt%), and methylcyclohexane (10-12 wt%). JP-9 was developed by the Air Force to meet its critical specifications for the fuels used in its Air Launched Cruise Missile. When subjected to the impact dispersion test, an intense fire resulted. No further work is planned with this fuel.

5. FRF-Type Fuels Augmented With RJ-5

FRF fuels and their fire-safety characteristics have been described in detail elsewhere.(8) Because they contain an appreciable amount of water, they suffer by having a reduced energy content. Since RJ-5 has shown the tendency to have fire-hindering properties, and also have a very high volumetric energy content (160,000 Btu/gal.), a blend of FRF and RJ-5 might result in a fuel that still has the fire-safety characteristics of FRF but at a much higher energy content.

Several months after the tests listed in Table 25 were completed, another series of impact dispersion tests were run using the same conditions as in the past. All the fuels containing RJ-5 that were tested in this series received a "D" rating in the impact dispersion test, and all containing water were microemulsions (Figure 7). These fuels are still stable micro-



FIGURE 7. RJ-5 MICROEMULSIONS

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emulsions to date (after 8 months) upon standing at room temperature. When only 3.0 vol% surfactant was used in attempting to prepare a microemulsion containing 5 vol% water and 5 vol% RJ-5 in 87 vol% DF-2, a trace amount of "cream" was visible at the bottom of the container after standing for several days at room temperature. When the surfactant content was increased to 3.5 vol%, the "cream" was eliminated and a stable microemulsion resulted.

In all cases, the RJ-5 increase in the volumetric ΔHc of the fuel compared to neat FRF without apparent sacrifice to its fire-hindering properates. Since RJ-5 appears to have good fire-resistant properties for exposure temperatures of 93°C (200°F) or less, it was not expected to have any deleterious effects under such exposure conditions when substituted in the FRF formulations. The results are summarized in Table 27. All the fuels, except No. 6, and including the FRF, were prepared using the same DF-2 (AFLRL Fuel No. AL-10115F, Cat 1-H) and all compositions are volumetric percentages.

IV. ENGINE STUDIES

A. CLR Engine Studies

1. Screening Studies

Prior to obtaining the Petter engine as a screening tool for high-energy fuel candidates, the screening tests were conducted using a CLR engine in the engine-testing facilities that were established and operated under a Department of Energy contract. Two of these fuel formulations, both related to the fire-resistant fuel program, are described below.

a. JP-10/H_O Microemulsion

FRF is a microemulsion consisting of 84 vol% DF-2, 10 vol% water, and 6 vol% surfactant. It has shown great promise as a fire-resistant fuel for the Army's application. It was reasoned that JP-10, with its relatively high

3	spositi	Composition, well		Lapact (1) Dispersion (1)	Density g/ml.15.56°C	Cetabe	Experimentel ABC (met).	Experimental Alic (net).	Percent Increase in The Volumetric Energy Content
Dan. No. DF-2 K	Ler erer	Surfactant	ž		(60°F)	2	Btu/lb	Btu/gal	Over Typical PL
1 84.0	7.5	6.0	2.5		0.8752	45	16,680	121,800	4.6
2 84.0	5.0	6.0	5.0	٩	0.8762	45	17,100	125,010	7.4
	0.0	6.0	5.0	۵	0.8856	64	16,105	119,000	2.2
	5.0	3.5	5.0		0.6730	47	17,290	125,920	8.2
5 91.5 S	5.0	3.5	10.0		0.8845	45	17,310	127,220	9.8
	i	8	50.0	۵	0.9657	[45] ⁽²⁾	17,900	144,220	23.9
7	10.0	6.0	ł		0 2:37	44	15.970	116,395	ł
Mase Fuel 100.0 -	1	ł	ł	(E)(3)	6.8504	53	18,665	132, 365	13.7

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A = No Pilot fimme emiargement.
B = Pilot fimme emiargement.
C = Pilot fimme dimensions less than doubled.
C = Pilot fimme totally obscured by transfent mist fireball.
E = Coalesced fireball with simultaneous pool buring.

All fuels listed in the table were prepared using a Cat I-H base fuel (AL-10115F) except No. 6 which used Code No. 8921. All experimental measurements made on Puel No. 6 were made or the same composition listed in brackets was made or a sample of the same composition with a category of AL-10115F bas a flash point of D0^oC (176^oF). The rating of "D" by the previous except dispersion test on Fuel No. 6 whold not of 71.7°C (161^oF); fuel AL-10115F bas a flash point of D0^oC (176^oF). The rating of "D" by the previous impact dispersion test on Fuel No. 6 should be applicable and comparable to the other tusis in the table and tested in this apparatus. ଞ

This fuel wes not tested by the impact dispersion apparatus. Value in brackets indicate a test on a fuel of similar composition. ĉ

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THE EFFECT OF RJ-5 SUBSTITUTION IN FRF-TYPE FUELS AND THE RESULTING PROPERTIES TABLE 27.

volumetric energy content [~142,000 Btu/gal. (net)] might be used to replace the DF-2 in the FRF and thus restore some or all of the energy loss caused by the water. Thus, the JP-10/H₂O microemulsion fuels were prepared to see if any fire-resistant properties might be imparted to this fuel. いちちちょうういろい

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Several different emulsifying agents were tried. Three microemulsion fuels of this type were prepared; all were stable for one year and one was still stable after two years when the stability test was terminated.

The terms Relative (Specific) Fuel Consumption (RFC), the Relative (Specific) Energy Consumption (REC), and the Relative (Specific) Volumetric Fuel Consumption (RVFC) are defined at the bottom of Table 28.

Screening tests on the latter JP-10/water microemulsion were conducted using the CLR engine. In Table 28, it can be seen that the REC values at 1000, 1500, and 2000 rpm are close to one, indicating that the thermal cycle efficiency when using this fuel is similar to that observed when the base fuel was used. A good comparison of the JP-10/water microemulsion fuel with DF-2 can be made using RVFC, i.e., the ratio of the BSVC of the test fuel to that of the base fuel. RVFC values less than unity are desirable as they would indicate better performance by the test fuel than the base fuel when using the same volume. The JP-10/water microemulsion has RVFC values close to unity over the speed and load range. This indicates a volumetric consumption rate very close to that of the baseline diesel fuel. Comparisons of this microemulsion with the base fuel can also be made using Figure 8a to 8f. It should be noted that the calculated net volumetric heat of combustion of this fuel (127,250 Btu/gal.) is almost that of a typical DF-2 (DF-2 control fuel = 128,465 Btu/gal.).

b. Carbon Black Slurries in FRF

Fire-resistant fuels in general are penalized in their energy content (Δ Hc) because of the water present. An attempt was made to reduce some of the loss in energy content by the addition of carbon. A 20 wt% carbon concentration in FRF was first prepared, but rheology problems suggested a lower

CLR ENGINE PERPORMANCE COMPARISON OF JP-10 MICROEMULSION (a) WITH DP-2^(b) TABLE 28.

KVPC(e)	1.060 1.068 1.060	1.057 1.040 1.061	0.936 0.98 6 1.104
REC(d)	1.056 1.056 1.047	1.055 1.366 1.051	0.925 0.914 1.083
Mc(c)	121.1 10.194 1.184	1.193 1.544 1.186	1.046 0.033 0.225
Thermal Lift, X	26.7 25.4 22.5	26.9 19.9 22.9	26.8 21 - 1
Specific Range hp-hr/ gal.	13.34 12.68 11.25	13.44 9.95 11.45	13.35 11.91 9.47
BSPC- Base Fuel Bal./ bhp-hr	0.750 0.074 0.084	0.070 0.073 0.082	0.080 0.085 0.096
BSVC- Test Fuel gal./	0.075 0.079 0.089	0.074 0.076 0.087	0.075 0.084 0.106
BSPC- Base Puel 1b/ bhp-hr	0.527 0.521 0.591	0.491 0.513 0.579	0.564 0.600 0.679
BSPC- Test Tuel 1b/ bhp-trr	0.591 0.622 0.700	0.586 0.597 0.688	0.590 0.662 0.832
Mase Puel Rate, 1b/m	2.70 2.00 1.52	4.02 3.12 2.32	4.72 3.93 3.17
Load, 1b/ft	24.0 16.9 11.4	24.0 13.8 11.8	21.0 15.6 10.0
Rack Position	Pull 3/4 1/2	Pull 3/4 1/2	Full 3/4 1/2
Bhp (obs)		6.85 5.23 3.37	
Bagine Speed,	1000 1000	1500 1500 1500	2000 2000 2000
Test No.			~~ ~ ~ ~

Density - 0.9463 g/ml at 15.56°C (60°F) (a) - JP-10/Water Mitroemulsion 32.73 wtZ, JP-10; 10.54 wtZ water; 6.20 wtZ surfactant; 0.50 wtZ cetane improver Heating Value (net) - 16,115 Btu/lb (calc), 127,250 Btu/gal.(calc)

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Density - 0.8438 g/ml at 15.56°C (60°P) (b) - DF-2 (Base fuel) Heating Value (net) 18,220 Btu/1b, 128,300 Btu/gal.

(c) - Relative (Specific) Puel Consumption (RFC)

RPC = BSPC_{test}/BSPC_{base}

(d) - Relative (Specific Energy Consumption) (REC)

REC - RFC (Heating Value_{net}Teat Fuel/Heating Value_{net}Base Fuel)

(e) - Relative (Specific) Volumetric Fuel Compution (RVPC)

RVFC = BSVC (Test Puel)/BSVC (Base Fuel)

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carbon concentration. The composition of the carbon/PRF slurry that was tested was 10 wt% carbon, 73.55 wt% DF-2, 10.32 wt% deionized water, and 6.13 wt% surfactant. The CLR engine performance comparison of this slurried fuel with base DF-2 is shown in Table 29. Figures 9a to 9f indicate that the suspended carbon caused a reduction in the performance compared to DF-2 over the three speeds and load ranges that were tested. Even at 10 percent carbon black, the viscosity of this test fuel was much higher than that of the base fuel and undoubtedly contributed to its poor performance by the degradation of the injection and atomization processes. As can be noted in the table, the REC's in all runs were greater than one. This was especially evident at the lower speed (1000 rpm). A THE PARTY AND A PARTY AND A

Figure 10 shows the REC, averaged over all loads, for each plunger diameter plotted versus the injection timing. Except for the 21-degree injection timing using the 9.5-mm plunger, the optimum plunger diameter appears to be the 8.5 mm. One surprising fact is the trend toward reduced energy consumption as the injection timing is retarded. If the carbon particles have long combustion time compared to the liquid fuel, it would be expected that increases in injection rate or advanced injection timings would improve the efficiency.

c. Catalyzed Carbon Black

As mentioned in Section III.D.2., the deposition of four different metals (manganese, iron, lead, and nickel) on carbon black was successfully achieved. Such metals might act as catalysts and aid in the combustion and utilization of the carbon black. The data in Table 30 compare the four "catalyzed" fuels with the base fuel diesel fuel, using REC as the criterion. The percentage of the metals coated on the carbon black listed in the table are those calculated assuming 100 percent of the metal was deposited in each case.

The REC values for the uncoated slurry, using the same injection timing and duration, are also listed in Table 30. Examination of this data indicates that the "catalyzed" carbon black slurries have improved the efficiency over the uncoated black slurry, especially with the nickel nitrate.

CLR ENCINE PERFORMANCE COMPARISON OF CARBON BLACK (10 wt2) SLURRIED IN FRF^(a) WITH DF-2^(b) TABLE 29.

KYPC ^(a)	1.511	1.290	1.237
	1.615	1.280	1.208
	1.682	1.137	1.767
BEC (4)	1.476	1.156	1.165
	1.519	1.186	1.110
	1.506	1.247	1.189
RPC(c)	607.1	46.1	446.1
	227.1	886.1	17801
	0 63. 1	994.1	178.1
Thermal Life, I	17.2	5.2	22.5
	16.8	1.2	21.8
	14.8	1.91	19.4
Specific Lange Lange Lange Lange Lange	8.14 7.96 6.9)1.02 10.46 9.04	10.65 10.76 9.16
NS WC	0.078	0.074	0.076
	0.078	0.075	0.079
	0.085	0.083	0.086
RSTC- Toot Pael Ma-hr	0.123 0.126 0.143	160.0 0.096 111.0	0.094 0.093 0.109
BSPC-	0.549	0.518	0.532
Naci	0.548	0.532	0.553
Naci	0.595	0.5e5	0.606
BSPC- Test Nest Nest Nest	0.935 0.960 1.089	0.691 0.728 0.842	0.715 0.70 8 0.831
Naes	3.49	4.50	5.69
Poel	2.85	3.45	4.50
Ib/a	2.28	2.79	3.67
16/64	19.6	22.8	20.9
	15.6	16.6	16.7
	11.0	11.6	11.6
Back Position	11M 2/1	7ull 3/4 1/2	7all 3/4 1/2
	2.97 2.97 2.09		
Rag I an Spaed.	1000 1000	1500 1500 1500	2000 2000 2000
ĔŔ		• • •	~ * *

Composition of Fuel that results is 73.55 wt% DF-2 (AL-8021F); 10 wt% Carbon Black; 10.32 wt% Delonized Water; 6.13 wtl Surfactant (64-8)

Density (Experimental) = 0.9144 g/ml at 1556°C (60°P) Calculated Beating Value (net) = 15,795 Btu/lb (120,530 Btu/gal.)

(b) - Df-2 (hase fue!) Heating Value (net) 18,220 Btu/1b (128,300 Btu/gal.) Density - 0.8438 g/ml at 1556°C(60°F)

- (c) Relative (Specific) Fuel Consumption (RFC)
- RPC BSPC_{test}/BSPC_{base}
- (d) Relative (Specific) Energy Consumption (REC)
- REC RFC (Beating Value_{net}Test Puel/Heating Value_{bet}Base Fuel)
- (e) Relative (Specific) Volumetric Puel Consumption (RVFC)
 - RVPC = BSVC (Test Peel)/FSVC (Base Puel)

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⁽a) - 10 wt% Carbon Black (Reven 1170) slurried in 90 wt% FWP



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FIGURE 10. REC AVERAGED OVER THE LOAD RANGE VERSUS INJECTION TIMING

	REC			
	Pull Load	3/4 Load	1/2 Load	
No Costing	0.99	1.03	0.99	
0.31% Manganese Nitrate	0.84	0.93	1.01	
0.1% Ferrocene	0.95	0.96	0.97	
0.05% Lead Acetate	0.97	0.99	1.12	
0.15% Nickel Nitrate	0.92	0.94	0.88	

TABLE 30. COMPARISON OF "CATALYZED" MOGUL L CARBON BLACK IN A 10 PERCENT SLURRY WITH BASE DIESEL FUEL USING REC VALUES AS THE CRITERION

B. Petter Engine

The majority of the candidate fuels for this program were liquid fuels blended with the engine baseline Cat 1-H. These blends also contained a cetane improver, if initial screening proved them to be of insufficient cetane quality to sustain normal engine operation.

A British-made Petter direct injection engine (Table 31) was used as a tool for comparing the fuel candidates against a baseline DF-2. A 60-cycle, 115/230-volt, 15-ampere, 3500-watt, AC generator was cradled and driven by a cog belt at twice engine speed to provide a load on the engine. The mechanical energy of the engine was converted to electrical energy in the generator, and dissipated as hest through immersion heaters in a drum of water. An electronic load cell and readout was attached, via a torque arm, to the cradled generator to be used as a measure of load (Figure 11).

The engine was governed to operate at 1800 rpm and a constant load of 10 lb in order to maintain a constant power test condition for all fuel blends. The barometric pressure, ambient temperature, and relative humidity were monitored for all fuel trials in order to correct engine performance for atmospheric conditions, as stated by the SAE diesel engine rating code (9). Appendix E is a listing of the data correction program used to reduce the engine data. The specific range (hp-hr/gal.), which is the inverse of the corrected volumetric fuel consumption, was the parameter upon which comparisons were made between the candidate high-energy fuels and the baseline Cat 1-H.



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IMMERSION HEATER IN WATER DRUM RHEOSTAT COG BELT RADLED LOAD CELL ENGINE AC GENERA-TOR 10.0 BURET FUEL MEASURED VOLUMETRICALLY (ml) DIGITAL LOAD READOUT (Ib)

FIGURE 11. PETTER SCREENING ENGINE INSTALLATION

The results from the screening procedures are presented in two different sets, due to a baseline drift after the Petter engine was idle for several months. The consistency of the results within each set of baseline data was very good, so there are no problems associated with the significance of the data collected. A comparison of the relative differences between two fuels that were examined in each baseline set allowed valid comparisons of the specific range based on the span of the 95-percent confidence intervals of the relative difference between these two fuels. The two fuels were Cat 1-H and FRF. 「「「「ないちをといれ」は後日

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1. Baseline No. 1

The results from the screening tests performed on the Petter engine are presented in two categories; one is high-energy fuel blends, and the other is the energy-augmented FRF blends. All corrected data are presented in Appendix B.

a. Fuel Blends

During the screening process, there were seven high-energy fuel candidates whose performances were examined with the Petter engine. Of the fuels blended, most ran very well in the engine, except for the naphthalene blends and the anthracene slurry. With the naphthalene blends, if the ambient temperature dropped below 20°C (68°F), the naphthalene would crystalize in the burret, and the engine would stall due to fuel starvation. The anthracene slurry proved to be too large in particle size to run through the Petter engine's gravity feed fuel system. Because of this, no data were available for comparison to baseline performance with the anthracene slurry.

Since the guidelines for a high-energy fuel require an approximate 10-percent increase in range over a typical DF-2, there were only two fuels which met this criteria. A blend of 50/50 vol% Cat 1-H and RJ-5 exhibited a 13.5-percent improvement in specific range (hp-hr/gal.) compared to Cat 1-H. The specific range of this blend was 16.77 \pm 0.18 hp-hr/gal., compared to a specific range of 14.78 \pm 0.1 hp-hr/gal. for the baseline Cat 1-H. The 95-

percent confidence intervals for the two fuels show that the comparison is statistically valid, with their deviation from the mean being 2.2 percent and 1.3 percent for the blend and Cat 1-H, respectively. Another candidate fuel which improved on baseline performance was a 75/24/1 wt% mixture of tetralin, Cat 1-H, and amyl nitrate. The calculated specific range of 16.16 \pm 0.01 hp-hr/gal. represents a 9.3-percent improvement over baseline performance with this blend. Due to cost consideration, and the performance evaluations, the tetralin fuel appears to be one of the more attractive high-energy fuel candidates. Table 32 and Figure 12 are comparisons of various high-energy fuel blends to the baseline.

b. Energy-Augmented FRF Blends

The energy augmentation of FRF to a performance level equal to the base Cat 1-H was considered an aspect of the High-Energy Fuels Program. There were three blends screened in the Petter engine, two of which exhibited a performance level comparable to the Cat 1-H. The third blend, which was an 85.6/14.4 wt% mixture of FRF and naphthalene, encountered the same operational difficulties associated with the other naphthalene blends. The performance increase of the FRF/naphthalene blend over the base FRF was insignificant in this study.

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The first of the two blends which displayed a considerable improvement over the base FRF was a 50/50 vol% mixture of FRF and JP-10. The increase in specific range was from 13.06 hp-hr/gal. for FRF to 14.71 \pm 0.01 hp-hr/gal. for the blend. This represents an 11.1-percent increase over base FRF, while falling a slight 0.5 percent from the Cat 1-H value of 14.78 \pm 0.1 hp-hr/gal. The second was a 63/37 vol% mixture of FRF and RJ-5, which displayed an 11.8-percent increase in specific range over base FRF. The specific range for this blend was 14.81 \pm 0.9 hp-hr/gal., which is actually a slight improvement over Cat 1-H at 14.78 \pm 0.1 hp-hr/gal. The comparison of the performance of these blends to baseline is statistically valid, due to the small span of the 95-percent confidence intervals for the specific range of these blends and baseline. Both of the blends performed extremely well in the Petter engine, without any engine modifications or operating

thermal Efficiency 28.5	28.6	29.5	27.7	27.9	28.5	28.4
 Change in Range Froe Baseline Cat 1-H 	13.6	٤.0	6.2	3.8	0.4	0.0
Span of 95% confidence Intervals as % Of Specific Range ± 0.65	± 1.1	± 0.05	t 0.80	± 2.40	± 0.50	± 2.45
Specific Range, hp-!.c/gal, 14.78	16.77	16.16	15.70	15.34	14.84	14.78
BSVC, gal./hp-hr 0.068	0.060	0.062	0.064	0.065	0.067	0.068
BSFC, 1b/hp-hr	0.483	0.484	0.498	0.520	0.486	0.493
	5.28 5.28	5.23	s. 12	s.12	5.26	5.15
Fuel Blend	Cat 1-H 50% Cat 1-K	50% RJ-5 75% Tetralin 24% Cat 1-H	1% Amyl Nitrate 99% JP-10	1% Amyl Nitrate 72% JP-10 ore No-bebolene	235 Augurungtene 35 Augyl Nitrate 90,993 DF-2	9.01% Phensnthrene 80.58% DF-2 19.42 Naphthalene

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All values represent data corrected for atmospheric conditions during the blend testing in the Petter Engine. The engine itself was run at constant power, i.e., constant speed and constant load.

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COMPARISON OF HIGH-ENERGY FUEL LANDIDATES TO CAT 1-H (BASELINE NO. 1) TABLE 32. ì

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FICURE 12. SPECIFIC RANGE OF VARIOUS ENERGY-AUGMENTED BLENDS COMPARED TO CAT 1-H (BASELINE NO. 1)

difficulties. The performance evaluations of the various energy-augmented FRF blends are listed in Table 33 along with the graphical results in Figure 13.

2. Baseline No. 2

E.

a. Fuel Blends

All corrected data for this baseline set are presented in Appendix F. During the screening process with this baseline set, nine separate highenergy fuel blend candidates were examined in the Petter engine. Of the nine fuel candidates, only one displayed an improvement of greater than 10 percent, while two others showed promise. The fuel which exhibited the largest increase in specific range for this baseline set was 99.6/0.4 wt% mixture of RJ-6M and amyl nitrate. The blend showed an increase in specific range of 11.9 percent at 17.15 ± 0.03 hp-hr/gal. compared to 15.32 ± 0.08 hp-hr/gal. for the Cat 1-H fuel. With this fuel, the operator noticed slight engine surging, although speed and load set points could be maintained. The surging may be due to the cetane number being below the engine manufacturer's specification of 45. The AFLRL experimental cetane number determined for this particular blend was 39. The potential of RJ-6M may be further realized with blend preparation based on increasing the cetane number.

One of the fuels that produced an increase in specific range, but fell below the 10-percent improvement level was a tetralin fuel. The blend was 97/3wt% combination of tetralin and amyl nitrate, which exhibited a 7.7-percent increase in specific range at a value of 16.50 ± 0.09 hp-hr/gal. Another tetralin blend screened in this baseline set, an 80/19/1 wt% mixture of tetralin/Cat 1-H/amyl nitrate, displayed only a 4.2-percent increase in specific range. As reported earlier, a 75/24/1 wt% mixture of tetralin/Cat 1-H/amyl nitrate showed a 9.3-percent increase over its respective baseline value. Table 34 and Figure 14 are a listing and visual representation of the anomaly that appeared in the data from the various tetralin blends. It is felt that this anomaly may be due to interrelated effects of the volu-

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INE NO.
(BASELIN
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BLENDS
FRF
TABLE 33. COMPARISON OF FRF AND ENERGY AUGMENTED FRF BLENDS TO CAT 1-H (BASELINE NO. 1)
DEN
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TABLE

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2 Thermal Efficiency	28.6	28.5	28.7	38 K	2	26.3	
X of Specific Mange From Cat 1-H	-11.6	-10.0	- 0.5	•	0.2	•	
X of Specific Range From Base FRF	ł	1.6	1.11		11.8	11.6	
Span of 95% confidence Intervals as % Of Specific Range	± 0.35	± 0.55	± 0.01		± 0.60	± 0.65	
Specific Range		13.33	:	14.71	14.81	14.78	
BSVC, gal-fuel/	hp-hr	710. 27 0		0.68	0.675	84 0	2
BSFC, 1b-fuel/	hp-hr	.558	100-	915.	538		184.
		5.17	5.15	5.27	5 24		5.10
	Fuel Blend	FLF.	85.6Z FRF + 14.4Z Napthalene	- Jul 1 05	507 JP-10	622 FLF + 372 RU-5	Cat 1-H

A All values represent data corrected for ermospheric conditions during the blend testing in the Petter engine. The engine itself was run at constant power, i.e., constant speed and constant load.

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FIGURE 13. SPECIFIC RANGE OF VARIOUS FRF ENERGY-AUGMENTED BLENDS COMPARED TO CAT 1-H (BASELINE NO. 1)

TABLE 34. RELATIVE VALUES OF SPECIFIC RANGE AND VOLUMETRIC HEAT OF COMBUSTION COMPARED TO CAT 1-H

Fuel	Blend	% Specific Range	Vol% Heat Of Combustion
Cat 1-H			
97% Tet:	ralin		
3% Amy	l Nitrate	7.7 ± 0.6	6.7
80% Tet:	ralin		
19% Cat	1-н		
1% Amy	l Nitrate	4.2 ± 1.05	6.2
75% Tet:	ralin		
24% Cat	1-н		
1% Amy	l Nitrate	9.3 ± 0.05	5.6

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FIGURE 14. COMPARISON OF CHANGE IN SPECIFIC RANGE TO CHANGE IN VOLUMETRIC HEAT OF COMBUSTION OF VARIOUS TETRALIN FUELS FROM CAT 1-H

metric heat of combustion and cetane number. The cetane numbers of these various blends had not been determined, but due to the chemical structure of the base tetralin, its cetane number is predicted to be around 10 or 12. The promise shown by the tetralin blend warrants further attention in developing its high-energy capabilities.

The other blend which showed improvement was a JP-10/naphthalene blend. A 77.6/19.4/3.0 wt% mixture of JP-10/naphthalene/amyl nitrate (AN) increased the specific range of the Petter direct injection engine by 7.1 percent. The value of the specific range was 16.41 ± 0.16 hp-hr/gal., versus 15.32 ± 0.08 hp-hr/gal. for the Cat 1-H baseline fuel. Initial problems in screening this fuel developed due to the extremely low cetane numbers of the basestock fuels; however, the addition of the amyl nitrate improved fuel performance adequately so that the test could be completed. The problems associated with all previous naphthalene blends were also evident in this high-energy fuel candidate. Table 35 and Figure 15 display the results for the various blends tested during this phase of the program.

Efflciency Z Thermal 29.5 29.1 29.8 29.5 29.0 29.1 28.9 28.7 28.6 28.9 I Change in Range Prom Baseline 11.9 7.7 7.1 5.4 8.4 4.2 2.3 0.6 0.3 ł Span of 95% Confidence Intervals as Z Cf Specific Range ± 0.45 ± 0.20 ± 0.60 ± 1.00 ± 0.25 ± 1.05 ± 0.40 + - 0.35 ± 0.50 - 0.55 Range, hp-hr/gal. Specific 15.32 17.15 16.50 16.41 16.15 15.97 15.67 16.05 15.36 15.41 gal/hp-hr BSVC , 0.0606 0.0623 0.065 0.058 0.061 0.062 0.063 0.064 0.065 0.065 1b/hp-hr **BSFC**, 0.462 0.487 0.484 0.492 0.498 0.504 0.496 0.490 0.499 0.476 BHP* 5.13 5.21 5.25 5.19 5.23 5.23 5.21 5.18 5.21 5.18 892 No. 6 Burner 011 102 Cat 1-H No. 6 Burner Cil JP-10 80% NO. 5 Burner 011 No. 5 Burner 011 3.0% Amyl Nitrate 19.4% Naphthalene 32 Amyl Nitrate Fuel Blend 12 Amyl Hitrate Amyl Nitrate 12 Amyl Nitrate 15% Naphthalene 97% Tetralin Tetralin Cat 1-H 77.62 JP-10 Cet 17 85% Cat 1-H 202 JP-10 Cat 1-B 3-2 791 201 802 192 202 12

All values represent data corrected for atmospheric conditions during the blend testing in the Petter engine. The engine itself was rum at a constant power, i.e., constant speed and constant load. *

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TABLE 35. COMPARISON OF HIGH-ENERGY FUEL CANDIDATES TO CAT-1H (BASELINE NO. 2)



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FIGURE 15. SPECIFIC RANGE OF VARIOUS ENERGY-AUCHENTED BLENDS COMPARED TO CAT 1-H (BASELINE NO. 2)

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b. Energy-Augmented FRF Blends

Further screening of energy-augmented FRF blends revealed two blends which showed parity with the baseline diesel fuel. The first blend exceeded the performance of the Cat 1-H by 2.6 percent based on specific range. This fuel was an 86.5/5/5/3.5 vol% blend of Cat 1-H/deionized H₂O/RJ-5/RA78 surfactant. The specific range for this fuel was 15.72 ± 0.08 hp-hr/gal., which is a 15.1-percent increase over the base FRF value of 13.41 ± 0.26 hp-hr/gal. For comparisons, the Cat 1-H has a specific range of $15.32 \pm$ 0.08 hp-hr/gal. Another energy-augmented blend which showed promise was 84/6/5/5 vol% mixture of Cat 1-H/surfactant/deionized H₂O/RJ-5. This combination of fuels increased the specific range of FRF by 10.4 percent to a value of 15.00 ± 0.04 hp-hr/gal., but fell short of baseline performance by a small 2.1 percent. Several other blends were tested, but results indicated relatively minor improvements. Table 36 and Figure 16 display the results for all augmented FRF blends examined.

V. CONCLUSIONS AND RECOMMENDATIONS

A. Conclusions

- Several of the screened candidates showed good potential for meeting the objective of improving range/power capabilities.
- The Petter engine has worked exceptionally well as a screening device.
- Several synthesized hydrocarbons, with cetane improver added, are feasible HEF candidates. These include RJ-5, RJ-6M, JP-10, and tetralin.
- A synthetically derived hydrocarbon, RJ-6M, improved the range of the screening engine by 11.9 percent over the baseline.
- The tetralin-based fuels are probably the most realistic highenergy fuels due to cost consideration and performance evaluations. The full potential of these fuels has probably not yet been realized, since cetane effects have not been fully evaluated.

Fuel Blend	BHP*	BSFC . 1b/hp-hr	BSVC, <u>Bal./hp-hr</u>	Specific Range, hp-hr/gal.	Span of 95% Confidence Intervals as a Percent of Specific Range	Percent Specific Range Prom Base PRF	Percent Specific Range From Baseline Cat 1-H	Percent Thermal Efficiency
FRF	5.21	0.544	0.075	13.41	± 1.9		-12.5	29.3
792 DF-2 102 (DI) H ₂ 0 62 EA78 52 RJ-5	5.25	0.543	0.073	13.67	± 0.5	1.7	-10.8	29.0
81X DF-2 10X RJ-5 5X H ₂ 0 4X EA78	5.19	0.507	0.068	14.67	± 0.25	8.3	- 4.2	28.9
84 X Cat 1-H 6 X EA78 5 X (DI) H ₂ 0 5 X RJ- 5	5.31	0.487	0.067	15.00	± 0.25	10.4	- 2.1	30.5
86.5% Cat l-H 5% (DI) H ₂ 0 5% RJ-5 3.5% EA78	5.24	ŋ.466	0.064	15.72	± 0.45	15.1	2.6	31.5
Cat 1-H	5.21	0.462	0.065	15.32	± 0.55	12.5	ł	29.5

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FIGURE 16. COMPARISON OF ENERGY-AUGMENTED FRF BLENDS TO CAT 1-H (BASELINE NO. 2)

 A 75/24/1 wt% blend of tetralin/Cat 1-H/amyl nitrate increased the specific range by 9.3 percent.

- Certain low-cost refinery products and streams have potential as high-energy fuel blending components, such as No. 6 Fuel Oil, furfural extract, and a polycyclic aromatic blending stock.
- The HEF and FRF programs have been brought closer together without diluting the effort of either.
- RJ-5, although costly, has an extremely high volumetric heat content and has shown fire-safety characteristics. It could be useful as a blending component.
- Blends have been prepared which show potential for high-energy, fire-resistant fuels.
- A 86.5/5/5/3.5 vol% blend of Cat 1-H, dionized H₂O, RJ-5, and EA78 surfactant improved upon FRF performance by 15.1 percent, which is a 2.6-percent improvement over the reference fuel.
- A 63/37 vol% mixture of FRF and RJ-5 increased the range of the base FRF to a parity with the reference fuel.
- A 50/50 vol% blend of Cat 1-H and RJ-5 displayed a 13.5-percent improvement over the reference diesel fuel (Cat 1-H).

B. Recommendations

- RJ-6M, one of the latest high-energy fuel candidates to emerge, shows great promise. Studies with this fuel should be vigorously pursued.
- Studies with the more promising fuel and fuel components should continue.
- Studies of the cetane effect on the performance of the high-energy fuels in the engine, especially with tetralin and its blends, should be initiated.
- Further work should be expended on developing blending criterion based on cetane number, due to inherent low cetane numbers of many of the basestock.
- All fuel blends which display improvements in range that meet the project guidelines should be further examined for performance optimization.

 Laboratory and engine studies with "catalyzed" carbon black should be pursued. HHAD

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- Some promising fuel and fuel blends have not yet been screened with the Petter engine. This should be done.
- Cognizance should be maintained with other high-energy fuel research, especially that conducted for the Air Force and Navy.
- Wherever feasible, efforts should continue to interface the HEF and FRF programs.

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APPENDIX A

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NO. 1 INSPECTION DATA ON SRC-11, MIDDLE DISTILLATE

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NO. 2 MATERIALS SAFETY DATA SHEET SEC-II MATERIAL

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APPENDIX A NO. 1

SRC-II Middle Distillate Inspections

Specific Gravity			Ash, w				0.0023
15.5°C/15.5°C		0.9840			Gum, mg/100 ml		25.1
Gravity, [•] API		12.3			d Stability, a		10.5
Viscosity					e Matter, mg/l		77.5
cSt at 37.8°C(100°F)	3.83			No., mg KOH/s	3m	0.31
cSt at 40°C(104°F)		3.68	Cetan				16
Flash Point, (D 93),	*C(*F)	80(176)			idue, 10% Btm	1, wt%	0.52
Aniline Point, *C(*F)	-9(16)	Water,				0.08
Pour Point, *C(*F)		-48(-54)			ip Corrosion,	50°C	
Cloud Point, °C(°F)		Too dark	•	2°F)			la
Hydrocarbon Type FIA	, vol%				f Combustion		
Aromatics		91.2	MJ/I	-			40.007
Saturates 🔪		8.1	Btu				17,200
Olefins		0.7	Visua:	-			
Elemental Analysis,	wt%				1500)		8.0
Carbon		86.15	Appea	rance		Dark gra	
Hydrogen		8.64	_			opaque.	Black
Oxygen		3.9	IR C	ILAG	Spectrum ind		
Nitrogen		0.82			of aromatic		
Sulfur		0.26			significant		
Ring Carbons, wt%					stitutions o		
Monoaromatic		30.5			rings. No o	rganic aci	.ds .
Diaromatic		22.6					
Triaromatic .		0.95					
Distillation	ASTM		GC Distr		on		
	<u>0</u>	<u>•</u> <u></u>	<u>• C</u>	<u>•</u> F			
IBP	187	368	151	304			
5% Recovered	204	400	187	368			
10	211	412	197	387			
15	216	420	210	410			
20	220	428	216	420			
30	228	442	232	450			
40	237	459	244	471			
50	245	473	258	497			
60	254	489	268	514			
70	262	504	280	536			
80	274	525	295	563			
90	289	553	311	592			
95	303	577	324	615			
EP	323	613	377	710			

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	MAIEKI	al Saf	ETY DATA SHEET		
RC II NATERIAL			Revised 8/79, Replace	15 4/79 	CRH
NUFACTURER'S NAME		DEU	IEMERGENCY	TELEPHONET	
The Pittsburg & Midw.	av Coal Min	ing Co	ſ	13) 226-101	1
Orth Port Levis, WA	98433		TRADE NAMES & SYNCHYMS		
EMICAL PAMILY			Naphtha PORMULA		
Coal derived aromatic S NUMBER	c naphtha		ND		
10			ND		
	SECTION	II - HAZ	ARDOUS INGREDIENTS	gel te da	
MATERIALS	*	TLV (Units)	MATERIALS	*	TLV (Units)
laphtha	99+	•	Hydrogen Sulfide	< 0.1	10 ppm
a pricing			Benzene	< 1.0	10 ppg
henols & Cresols No information is on otential of this par ssions, associated up roblems. Therefore inimum. See Section	rticular mai	terial. He iquefaction mended th	the long-term toxicity and, powever, the health hazards, a processes are recognized a hat inhalation and skin expo	or carcin particula	ogenic rly skin al health
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can occur from	the possi	lble sk	in	absorption,	inhala	tion, and	ingesti	on of this material
due to the pre Emergency and P	sence of 1	ohenoli	c	compounds.			-	
sing with cast	or or oliv	ne oil	fo	r large skin :	lurfac	e exposure	S. TYE	CONTACT-Flush immed.
DO NOT INDUCE	VONITING ((aspira	۱.	hasard) Give	L-2 oz	act. chas	ccoal fo	. if necessary. INGES: llowed by 1-2 glasses cases of contact.
				TION VI - RE		ITY DATA		
STADILITY UNSTA	Materials to and	OLE X						
Reacts HAZAROOUS OLCOW When hea	with stro		41	zing material	·			
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DEVACUATE A	REA		PH	ATORY PROTECTIC		SUNTS IN HO	38	DAWAY WITH WATER
STOP FLOW			1	ROTECTION R SECTION VIII)		WARDLED CO	ADITIONS	DESERVE ODVERNMENTAL
DELIMINATE A	LL SOURCES	. 🚺 AÐ	604	B OR SCRAPE UP		INERATE VIL		A REMOVE SOLLED CLOTHING
AVOID INHAL	ATION		CVI	UM UP		AS LANDFIL	E AREA O	A D OTHER
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		DUR		NORMAL USE	500 Excl	CASES YAP	1874	Process in the provide
GENERAL VENTILATION YOS YOS								
LOCAL EXHAUST Yes Yes								
RESPIRATORY PRO	ECTION (1-3)				2,	if needed		3, large spills
1. Particle - Removing A (Mechanical Pilio		Respirator		2. Ges and Vapor-Ros Respiretor (Ca		Purifying I		Noch Positive Prostare - Domand Supplied Air
EYE PROTECTION	SAPETY OL	4685	X	CHEMICAL GOOD	LES X		.0	(E) EXCELLENT
PROTECTIVE	NEOPRENE		9	POLYVINYL ALEO	HOL MR	POLVETHYL	ENL	WR (F) PAIR (F) POOR
GLOVES	NATURAL R		7	BUTYL AUBBER	MR	POLYVINYL		WR (NR) NOT RECOMMENDED
CTALA PACTECTIVE	WOTK 18 1	192-11	R	idation of a d	kin b	arrier or	IAN DEIG	Se work and several
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The data and recomi		Noricia	-	in are based stores a		TOR THE		RTATION SPILLS ON LEAK
and the research of	uthers, and a	ve believ	N,	iu be annuele. He	parent	er Chem	TRBC - 6	00-424-9300
of their accuracy is i without worranty, ex his own determinati	press or implie	d, and th	i p	mon receiving them	shall me	ke CENTI		INSPORTATION EMERGENCY

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U.S. DEPARTMENT OF LABOR Occupational Safety and Health Administration MATERIAL SAFETY DATA SHEET

SRC II MATERIAL				Revised 8/	9, Keplaces	7/79 CRH
5. (*) (*) (*) (*) (*) (*) (*) (*	14. a ti	14 36	SEC	TION I	the second	
MANUFACTURER'S NAME The Pittsburg & Hide		al Mini	ing Co.	EWENS!	(713) 226-10	-
ADDRESS (NUMBER, STRFLT, C	ity,st/	ATE & ZIP	CODE)			
NOTE FOTE LEWIS, WA CHEMICAL NAME & STNONTH	<u>yea</u> 3			TRADE NAMES & STANDATAS		
NA CHEMICAL PAMILY				Heavy Distill		
Coal derived heavy d	isti)	late		TISCA REGISTATERTAT		
ND				ND		
A PARTIN IN THE SUCCESS		ECTION	II - HAZA	RDOUS INGREDIENTS	a sacally far	a retraining
MATERIALS		%	TLV (Units)	MATERIALS	*	TLV (Units)
Heavy Distillate		100	•			1
						<u> </u>
recommended that in recommended precaut	halat ions	ion and be obse	i skin expo prved when	th problems. Therefore sure be kept to a minimu- handling this material.	an and that a	11
ويستبته محتفظة معتبد ومسيعه فتباد			regulatio			
	y	SECT	TION III -		her the first	Service States
BOILING POINT C (F)		288-454		18.4 / 18.4 C		
VAPOR PRESSURE (MM HS.)		ND		PERCENT, VOLATILE BY VOLUME (%)		
VAROR DENSITY (Arras)		MD		EVAPORATION MATE	ND	الوراد بيرين متحصيل
BOLUBILITY IN WATER		Negligi	ble			او انو منذ است.
APPEARANCE AND ODOR				, oily odor.		الماليب والمريد والمريد
	CTION	1.11/		EXPLOSION HAZARD DAT	A 1.7 He	
			A REAL PROPERTY AND ADDRESS OF AD	TELAMMABLE LIMITE		UL
EXTINGUISHING MEDIA	(250)	1)_010	eed cup			
	ARBON	DIORIDE	DAV	GHEMICAL 🖸 FOAM	WATER 1	IPRAY (FOG)
SPECIAL FIRE FIGHTING PRO flush spills away f equipment for enclo	rom f:	ire exp	osure, and	to keep fire-exposed c to disperse vapors. Us al sinks in water.	ontainers co e air-suppli	ol, ed rescue
Jikely to give a sparke.	0	ombusti Dient v	bla. Heat: apors for i	ng this material to its gnition upon exposure t	flash point o flame or i	is noendiary

NA + Not Applicable

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ND - No Data Available

Gulf Medilies Form OSHA-20

and the second s

						HEAVY DISTILLATE
	25 A. 42		TION V - HEAL		27.	
THREE-OLD LIMIT	ALUE Sugg	sated o	uideline for ai	rborne en	posure to con Section II.	al derived materials:
			ves. Paren sektastade eter			MACT WITH THE SILIN. A TROUGH THE
-						and upon marting seculing moleco. (Altical: Natur pressure skim indiction —
NY CLUP OF CHURCH I CITRELY PAIRLO	I'M PAR, SILICTO	ns. Jacadra		t fillet; within	A PRY HOURS TENDE H	LL MOPE BOLLIN, DISCLORE, MO
						EVE CONTACT ; Fuille with carrous
						(38 SILECTION: Grandiery sedical nes for theatment of this type of
1040 (1.E., HE1699), (P NEXETIC TIME, NO VE	no antasing).		
en an	0.*** <u>*</u> ******		ECTION VI - RE	ACTIVITY		
STABILITY: UNSTA			CONDITIONS TO AVOI	5	¹	
INCOMPATABILITY (Aaterials to avo		ts with strong o	ridizina	materials	
HATANSOUS DECOM	TO DECORD		n, it will emit			fumes.
HAZARDOUS POLYMERIZATION:	MAY OCCUR	WIL	LNOT OCCUR X	INCITIONS 1	AVOID	
A. S. Sparster		SECTIO	N VII - SPILL O	R LEAK	PROCEDURES	
			PIRATORY PROTECTIO			DAWAY WITH WATER
Stop PLOW			PER SECTION VIN)		ATE UNDER	
ELIMINATE A	L SOURCES		IONE OR SCRAPE UP		ATE UTING AFTE	A BREMOVE SOLLED CLOTHING
AVOID INHAL		_			LANDFILL	
DAVOID DEAMA	L CONTACT					
a ginala	SEC 1	TION V	/III - SPECIAL P	BOTECTIC		ION LO CONTRACTOR
						16. 2102 0.1
			ING NORMAL USE	PX2116	NG YLY	APPLICAL (E.G. THE RMAL APPLICATIONS)
GENERAL VENTILA	TION		Yes	Y	••	
LOSAL EXHAUST		Pr	eferred	Y	••	
RESPIRATORY PROT	ECTION (1-3)			1, if		
1 Periick - Removing A (Mechanical Filie		Respiretor	2. Ges and Vepor - Rei Respirator (Ca			e Mask Podsive Pressure - Demend pe Supplied Air
EVE PROTECTION	SAPETY OL	48865	X CHENICAL BOOG	LES PI		IEI EXCELLENT
PROTECTIVE	NEOPRENE		G POLYVINYL ALEG	HOL WR M		HR (F) PAIR (F) FOOR
GLOVES	NATURAL A		BUTYL RUBBER		LANINAL CHIORI	E MR ///A) NOT RECONMENDED
		The a	pplication of a everal times du	ring work	Is recommend	ed .
	14 M	41	TION IX - SPEC			in the in the second start
of mists, fune			S AND STORING AVO: bintain good ver	ntilation	nd sys contact. Fresh chan	e of work clothes
daily. Shower	ing and gl	othes	change recomen	ded at th	e and of eact	shift. Combustible drums with vater befor
hydrocarbons, (IS BAY	CONTAIN COMPUSE	1916 Vapo		discardine,
		data	suggest that SRG	. materia	is have the p	otential of producing
	fetoto	xic ef	focto. Therefor	re, it is	prudent to p	revent fertile females
L		NOTICI	ng significant (ADOSUTE		SEL. DRTATION BPILLE OR LEAR
The data and recom	nundetkins p	esented l	, wrein ere bewe up um u wit tu be ewurete . M	we research	EMEROENCIE	
of their occuracy is i	nade, howeve	r, and the	products discussed are	distributed	CHEMICAL T	RANSPORTATION ENERGENCY
			ne person receiving them Thereof for his perticul		CENTER).	

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1	cupational	Safety a	MENT OF LAB nd Health Adm ETY DATA	inistratio	n			
SRC II NATERIAL			Re	vised 8/79.	Replaces	7/79 CRH		
		SEC	TION I SALE			110.18		
The Pittsburg & Midwa	y Coal Mini	ng Co.		TEMENSENCY Y	2) 226-101	1		
ADDRESS INUMBER STREET, C North Fort Lewis, WA	114 STATE & 21F	CODE)						
CHEMICAL NAME & SVHONVM	1		TRADE NAMES I E	VNONVME Distillate		ليخيرون حداد مرحد ال		
Cosl derived middle d	listillate		VORMULA KA					
CAS HUMBER YEAR NEW YEAR NOT THE STATE OF TH								
SECTION II - HAZARDOUS INGREDIENTS								
MATERIALS	*	TLV (Units)	MATER		*	TLV (Units)		
Middle Distillate	99+	*	Bensene		<1	10 ppm		
*The health hezerds,								
processes, Are recon recommended that in recommended precaut: DOT HAZARD CLASS: (halation and	skin expo rved when 1	sure be kept to a	minimum and	i that al	1		
		ار بنور هي خلك الجاكو اور هي بين الي اي ا		a an an	u. Maranti	9. 191 - data		
BOILING POINT C (T)	SEC	TION III ~	PHYSICAL DATA		i Sulla			
	177-288°C		15.5 / 15.6 C	,	0.97			
VAPOR PRESSURE (mm HL)	10	L	PEACENT, VOLATILE		97			
VAPOR DENSITY (Ar +))	10		EVAPORATION RATE		10			
SOLUBILITY IN WATER	Slightly s	oluble						
			trong creosoteli	ke odor				
	ينوحد بسبة ويناهده والم				2. <u>)</u>	1110.102 S. Con		
LASH POINT	CTION IV -	PIRE AND	EXPLOSION HAZA					
ERVINGUISHING MEDIA	CARBON DIGXIDE		CHEMICAL					
SPECIAL FIRE FISHTING PRO Spills sway from fir ment for enclosed as	re exposures	and to di	sperse vapors. U	ise alt-suppl	1 05 705 0	1, flush us squip-		
FIRE AND EXPLOSION HAYAR Bither misting of the to give sufficient var	Combustib material th pors for ign	rough hand ition upon	ability similar t ling or heating t exposure to flam	o its flash w or incendi	point is ary spar	likely ks.		
NA + Not Applicable	ND . No Dete	Available		Quil Me	dilled Fac	M OSHA-20		

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MIDDLE DISTILLATE

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		SE	CTI	ON V - HEAL	TH P	IAZ	ARD DATA		
THREEMOLD LIMIT	as bens	ested ene sc		les. See Sec	tion	"u	exposure to com	de	FIVED BECOFILIES
	ADDORP., INDALAT	ial, AB I NGN POW		. OF thes extended and CUTION: these prese	TO THE		net of Indialic Contains. Figh HW COUR with Holking	ha	NG & MEAN, MARINE, CHI GODA Digita & Manazian Bandaras Any Alia, Multitude, Bullary Mill
DE ORACT: FLIDA DADA (Administrati Hadada), Era	. WITH WATER POR E 1-2 OF ACT. CH	15 mm. 1 MCDA, POL		17101: North Mark Data 1917 1-2 Glasses of Hill	NNE, SI K CH 24	NE AN	es de quive delle Geranne e	jiesi Higi	UN LUNE BUD DATAS OFODAS. 101: 10.000 MLC VIIIIG ND. AN POLAL CARS OF COM-
							C TISRE, & HOAD STESSING		
STABILITY: UNSTA		BLE X	_	TION VI - RE			Y DATA	des er.	
INCOMPATABILITY (Materials to ave	Head	ts:	with strong of	xidi	zín	g materials.		
When heated	to decomp	OBUCTS	m,	it will emit	irri	tat	ing and toxic fo	-	•
HAZARDOUS	MAY OCCUR	w1		OT OCCUA X C	ONDIT		TO AVOID		
land a start from		SECTIO)N	VII - SPILL O	RL	EAK	PROCEDURES		
DEVACUATE A	REA		ispi S Pi	ATORY PROTECTIC	20%	MOU	MATE SMALL		WAY WITH WATER
STOP FLOW			s're	A SECTION VIII)	5	ic ini	ROLLED CONDITIONS		ALL & WATER QUALITY
DELIMINATE A	LL SOURCES	, 🖾 AI	SOR	B OR SCRAPE UP			ERATE USING AFTER	6	EMOVE SOILED CLOTHING
AVOID INHAL	ATION	x v	NCUI	un up		XEX	IN REMOTE AREA O	0	THER
AVOID DERM	AL CONTACT								
	SEC.	TION	VII	- SPECIAL P	ROT	CT)N	
		ø	AING	NORMAL USE	222	Î	AFER YAMORS	1.56	IAL (S.C. THERMAL CERTING ANDAL LICATIONS)
GENERAL VENTILAT			Y	14			Yas		
LOCAL EXHAUST			Y	**			Yes		
RESPIRATORY PROT	ECTION (1-3)					2,	if needed		, for large spills
I Periscie - Removing A (Nechanical Filie		Rophilo	, 	2. Gas and Vapor-Rei Respirator (Ca		A# N			editor Pressure - Demand Wed Air
EYE PROTECTION	SAFETY OL		X	CHEMICAL GOOD	LES	X			(E) EXCELLENT (C) GOOD
PROTECTIVE GLOVES	NEOPHEN:		٥	POLYVINYL ALCO		-+	POLYETHYLENE	XR	(F) FAIR (F) FOOR (NR) NOT RECOMMENDED
OTHER PROTECTIVE	NATURAL A		? ml	CATION OF A	-		POLYVINYL GHLORIDE PISY CYSEM DETO		
			1	times during	rork	11	racomended,		
PALLAUTIONS TO B	AL ANT	A SE	CTI	ON IX - SPEC	IAL	PRE	CAUTIONS		old the inhalation
of mists, fum Showering and	es, or vaj clothes (av contai)	bors. Change	na Te	intain good vo commended at (ontij the e	ati md	of each shift.	ic a Comb	work clothes dail; ustible bydrocarbon before discarding.
							the potential of fortile female		Mucing Setetotit
				this material		- UH 1			··- ····
The date and recomm	ne nde thans p i	NOTICE	here	in one based upon a		ner: h	EMERGENCIES,	CAL	-
and the rescarch up	uthers, and a	he beile		in be occurate. No	o pue n	m/ee	CHEMTAEC - I	10-(134 - 9300

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of their accuracy is made; however, and the analysis discussed are distributed without warranty, express or implied, and the person receiving them shall make his own determination of the suitability thereof for his preticular purpose.

(CHEMICAL TRANSPORTATION ENERGENCY CENTER).

APPENDIX B

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NO. 1 AFLEL FUEL NO. AL-10115-F (CAT-1H) NO. 2 AFLEL FUEL NO. AL-8821-F (DF-2) NO. 3 AFLEL FUEL NO. AL-7225-F

APPENDIX B, NO.1 AFLRL FUEL NO. AL-10115-F(CAT-1H)

API Gravity, 15.56°C (60°F)	34.9
Flash Point, Pensky-Martin Closed Cup, °C (°F)	80 (176)
Pour Point, °C (°F)	-16 (+3.2)
Cloud Point, °C (°F)	-4 (+24.8)
Carbon Residue, X	0.14
Coppor Corrosion	14
Ash,X	0
Lamp Sulfur, X	0.399
Viscosity, 37.8°C (100°F), cSt	3.21
,40.0°C (104°F), cSt	3.12
Saturates, X	71.5
Aromatics, %	28.5
Cetane Number	53
Hydrogen, wt%	13.09
Carbon, wt%	86.46
Nitrogen, ppm	36.5
Net Heat of Combustion, Btu/1b	18,665
H ₂ 0, %	0
Sediment, X	0
TAN	0.03

Aniline Point, *C (*F)

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67.7 (153.9)

APPENDIX B, NO. 2

AFLRL FUEL NO. AL 8821-F

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API Gravity, 15.56°C (60°F)	35.2
Density, g/ml 15.56°C (60°F)	0.8484
- · · ·	
Carbon, wt%	86.66
Hydrogen, wt%	13.34
Flash Point, Pensky-Martin Closed Cup, [°] C (°F)	71.7 (161)
Fire Point, °C (°F)	84 (183.2)
Viscosity, 40°C (104°P), cSt	3.2
Cetane Number	51.0
Sulfur, vt%	0.47
Copper Corrosion (D-130)	1A
Net Heat of combustion, Btu/1b	18,450
Cloud Point °C (°F)	-1.0 (+30.2)
Pour Point °C (°F)	-10.0 (+14)
Accelerated Stability (D-2274), mg/100 ml	2.7
TAN	0.03
Stream Jet Gum mg/100 ml	3.2
Saturates (FIA), vol%	69.1
Olefins (FIA), vol%	1.5
Aromatics (FIA), vol%	29.4
K.F. Water, %	0.03

B-4

APPENDIX B, NO. 3 REFERENCE GRADE DIESEL FUEL, #AL-7225-F [MIL-F-46162A(MR)]

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API 15.56°C (60°F)	36.1
Flash Point, Pensky-Martin Closed Cup *C (°F)	60 (140)
Cloud Point, °C (°F)	-21(-5)
Pour Point, °C (°F)	-24 (-11)
Viscosity, 40°C (104°F), cSt	2.07
37.8°C (100°F), cSt	2.17
Carbon Residue on 10% Bottoms, wt%	0.15
Lamp Sulfur, wt%	0.35
Sulfur (by X-ray) %	0.31
Copper Strip Corrosion	1A
Ash, wt%	0.0009
Water and Sediment, vol%	0.01
TAN	0.01
Cetane Number	48
Saturates, (FIA), vol%	70.9
Olefins, (FIA), vol%	1.6
Aromatics, (FIA), vol%	27.5
Density, 15°C g/ml	0.8438
Gross Heat of Combustion, Btu/1b	19,450

APPENDIX B, NO. 3 (Cont'd)

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Boiling Point Distribution, C

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IBP	167	89
10% evap'd	218	211
20%	228	220
30%	233	232
40%	238	238
50%	242	251
60 X	248	258
70%	257	271
801	270	287
902	293	312
95%	313	333
E.P.	336	393
Recovered, %	99	
Residue, %	1	
Loss, %	0	

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FREZING TUBE TECHNIQUE

APPENDIX C

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C-1

APPENDIX C

FREEZING TUBE TECHNIQUE (a) (b)

In this AFLRL-developed method, the prepared samples of known concentration were placed in plastic tubes (4.5-cm OD, 3.8-cm ID, 0.32-cm wall, and approximately 31 cm in length) which were plugged at the bottom with a No. 8 rubber stopper (Figure a). Almost all the data were taken at room temperature; however, a controlled temperature bath could be used for future samples if settling rates at other temperatures are desired (Figure b). After the desired time period for settling, the tubes were frozen at approximately - $80^{\circ}C$ (-112°F) (Figure c), and each tube cut into six segments (Figures d and e) and measured by length. After each segment has thawed, it was thoroughly stirred to give a uniform sample. A small 5- to 10-gram sample was removed and then analyzed for solids (carbon) by a modified ASTM D 893-78 procedure, the standard test for insolubles in (used) lubricating oils. The procedure, as modified, consists of weighing the sample in a centrifuge tube, washing it three times with 75 to 100 ml of pentane per wash, and removing the pentane by decanting after centrifugation. The sample is dried in an oven at 104°C (220°F) for at least 30 minutes, cooled in a desiccator, and then weighed.

Although a good method, it is time consuming and it tends to give slightly high values for the recovered carbon. Although the values appear to be high, they are correct relative to each other, a phenomenon whch might be caused by the incomplete removal of the last trace of diesel fuel adsorbed by the carbon black. Although the method is not yet quantitative, it is an excellent indicator of stability, since all samples are prepared in an identical fashion.

- (a) From Reference 3
- (b) For revision to this technique, see this report, III-D-lb(3)(a).

C-3



a. Tube Containing the Carbon Slurry



FREEZING TUBE METHOD FOR MEASURING CARBON SEDIMENTATION



d. Frozen Tube Just Prior to Slicing



e. The Six Resulting Segments After Slicing

FREEZING TUBE METHOD FOR MEASURING CARBON SEDIMENTATION (CONT'D)

APPENDIX D

IMPACT DISPERSION TEST

Program Title: Sponeor:

Contract No.:

Development of Impact Dispersion Fluid Flammability Test

U.S. Army

DAAD05-67-C-0354/DAAD05-70-C-0250/DAAK-73-C-0221/ DAAG53-76-C-0003/DAAK70-78-C-000;

SwRI Project No.:

Start/Complete Dates:

Reports or Publications:

10-2078/10-2798/10-3630/10-4296/10-5070

1969/1978 Continuing Refinement of Facility and Procedures

Weatherford, W. D., Jr., and Schaekel, F. W., "Emulsified Fuels and Aircraft Safety," AGARD/NATO 37th PEP Meeting "Aircraft Fuels, Lubricants, and Fire Safety," The Hague, Netherlands, May 1971 (AGARD-CP-84-71, pp. 21, 1-21, 12).

Weatherford, W. D., Jr., and Wright, B. R., "Status of Research on Antimist Aircraft Turbine Engine Fuels in the United States," AGARD/NATO 45th PEP Meeting "Aircraft Fire Safety," Rome, Italy, April 1975 (AGARD-CP-166, pp. 2, 1-2, 12, DDC No. AD-A011341).

Weatherford, W. D. Jr., and Schaekel, F. W., "U.S. Army Helicopter Modified Fuel Development Program—Review of Emulsified and Gelled Fuel Studies," prepared by Southwest Research Institute, U.S. Army Fuels and Lubricants Research Laboratory, under U.S. Army Contract No. DAAK02-73-C-0221, AFLRL No. 69, June 1975 (DDC No. AD-A023848).

PROGRAM SYNOPSIS

Technical Objectives: Develop a repeatable bench-scale laboratory technique for assessing fire-safety characteristics of various flammable liquids.

Appro: th: Impact dispersion experiments are conducted in a well ventilated enclosed facility developed for this purpose. These tests involve allowing a 2-liter glass vessel, containing about 1.2 kg of fuel, to fall freely 6 meters onto a steel target plate, originally embedded in concrete and surrounded on two sides by gas pilot flames. In the present facility, the target plate comprises a horizontal, elevated 2.5 cm-thick steel plate with electric surface heaters attached to its under side so that its surface temperature can be adjusted and controlled independently.

The relatively low vertical velocity of 11 meters per second developed by the glass vessel during this free fall corresponds to total occupant survivability during a vertical helicopter crash, but it is near the onset of marginal survivability. The glass containers are filled to an ullage of about 2 percent of the total volume for each test. A television camera (with zoom lens) is located about 6 meters from the impact point, and this is used to document the test results on video tape. A background grid provides a dimensional frame of reference, and subsequent examination of the video tape by slow motion (and stop action) provides reduced data. Tests are conducted at several different temperature levels, from about 25° to 99° by preheating the fuel sample and the steel target plate to the desired temperatures.

Accomplishments: The following data reduction system placed the impact dispersion results on a semiquantitative basis. A rating from "A" through "E" is assigned to the results of each experiment, depending upon the observed flaminability characteristics. These range from "no pilot flame enlargement" for highly effective antimist fuels through "pilot dimensions less than doubled," "pilot flames totally obscured by transient mist fireball" (neat, low-volatility fuels), to "coalesced fireball with simultaneous pool burning" (volatile liquid fuels). This method of quantifying the results of impact dispersion tests has proved usable and correlatable with other experimentally measured flammability properties.



Heated Impact Pad, Gas Pilot Array, and Fluid Container With Solonoid Release (Lowered for Photograph)



Peak Splash With No Fireball (Typical of Antimist Fuels)



Peak Mist Fireball

D-3

APPENDIX E

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DATA CORRECTION PROGRAM FOR PETTER ENGINE DATA

2 #FILES (0,5) 3 C* ************ 4 C 5 C PEDRP-Petter Engine Data Reduction Program, a program used to correct performance 6 0 7 0 parameters from the PETTER screening 8 C engine for atmospheric conditions. The 9 0 mean and 95% confidence intervals are 10 0 then calculated for corrected parameters. 11 0 12 0+ *********** 13 0 PROGRAM FEDRP 14 15 DIMENSION A(100,7), B(100,5), C(30), D(5) 16 DIMENSION NAST(2), LAST(25), NFAC(3), NSRG(3), NAST(25) 17 0 18 0 19.0 OPEN FILE "ATM" WHICH CONTAINS THE SHE ATMOSPHERIC CORRECTION 20 0 FACTORS. 21 C 22 0 BD = DRY ATMOSPHERIC PRESSURE .13 C ⇒ AMBIENT AIR TEMP.(F) T 24 0 AB . CONVERSION TO ABSOLUTE (EMP.(R) 25 0 26 0 27 OPEN(501,FILE= ATM) 28 READ(501,*)BD,T,AB 29 F=T+AB 30 C 31 0 32 C OPEN FILE (BTU) WHICH CONTAINS THE NET HEAT OF COMBUSTION OF 33 C THE FUEL BLENDS EVALUATED WITH THE PETTER ENGINE. 34 C 35 0 OPEN(504 FILE= STU() 36 37 C 38 C 39 LU=1 48 450 URITE(LU,1) FORMAT(12, 'ENTER LU NUMBER FOR PRINTER----() 41 1 42 READ(LU,+)LP 43 L=7 J≈5 .14 - 5 HCJ=J*6 NBF = 100 46 47 (48 C 49 C THE SPECIFIC RANGE FILE IS WERE THE CALCULATIONS FOR 50 Ç SPECIFIC RANGE ARE PLACED IN ORDER TO COMPARE FUELS. 51 0 52 0 33 WR170(10,200) 54 200 FORMATCIN, SPECIFIC RANGE FILE HHMF-----55 READ(LU, 201)NBRG 36 201 FORMAT(3A2)

Sub-Survey of States

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E-3

57 C 58 C FUELS CAN BE ACCESSED INDIVIDUALLY OR ACCESSED FROM 59 C 60 C A FUEL ACCOUNTING FILE FOR A SPECIFIC BASELINE SET. 61 C 62 C 63 WRITE(LU,2) 64 2 FORMATCIN, 'A SPECIFIC FUEL FILE ? Y OR N () READCLU, 105 HER 65 66 105 FORMAT(1A1) 67 C 68 C 69 IF(LFN.NE.1HY.AND.LFN.NE.1HN)G0 TO 30 70 IF(LFN.NE.1HY)GO TO 8 71 C 72 C 73 C ACCESS FUELS INDIVIDUALLY 74 C 75 C 76 WRITE(LU,3) 77 3 FORMAT(1X, 'INPUT FUEL FILE NAME -----> 78 READ(LU,5)NAST 79 5 FORMAT(2A2) 80 OPEN(502, FILE=NHST) 81 WRITE(LU,6) FORMATCIN, 'INPUT FUEL BLEND NUMBER-----82 6 83 READ(LU,+)IFNO 64 C 85 C SUBROUTINE FEND CORRELATES FUEL NUMBER WITH NET HEAT OF 86 C COMBUSTION. 87 C 88 C 89 C 90 CALL FFNO(IFNO, 504, BTLB, LAST, *23) 91 C 92 C GO TO 13 93 94 8 IFCLEN, EQ. 1HN, AND, NUT, EQ. 1HY DREWIND 503 95 C 96 C 97 C FUELS ARE ACCESSED THROUGH A FUEL HOCOUNTING FILE WHICH 98 C CORRELATES WITH A SET OF BASELINE DATH. 99 C 100 C WRITE(LU,9) 101 FORMATCIX, 'NAME OF FUEL ACCOUNTING FILE------102 9 103 READ(LU, 10)NFAC 104 10 FORMAT(3A2) 105 OPEN(503,FILE=NFAC) 106 LEG=0 107 11 READ(503, +, END=300) 108 LEG=LEC+1 109 GO TO 11 110 300 REWIND 503 111 DO 21 LL=1, LEG

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READ(503,12)NAST, IFNO
112
113 12
           FORMAT(2A2,1X,13)
114
           OPEN(502, FILE=NAST)
115
           CALL FFNOKIFNO, 504, BTLB, LAST, #230
116 0
117 0
118 0
        PRINT OUT HEADER
119 0
120 C
121 13
           WRITE(LP, 14)LAST
           FORMATCIHI,40X, PROJECT NO. 05-5957-130 JAZ.30N, CORRECTED JIN,
122 14
          1 PETTER ENGINE DATA FOR FUEL BLENDS 1,22,238, FUEL BLEND: 1,18, 225A2,22,228,60(1+1),2,248, FUEL ,68, 18HP1,78, 18SFC1,68, 18SVC1,
123
124
          35%, "SPECIFIC", 3%, "THERMAL", 2,65%, "RAHGE", 2,24%, "NO. 1,15%, 4"LBZHP-HR", 3%, "GALZHP-HR", 3%, "HP-HR"/GAL", 4%, "EFF, , 2,22%, 60% (#"))
125
126
127 0
128 0
        DETERMINE NUMBER OF ENTRIES IN A FUEL FILE
129 0
130 C
131 0
           N = 0
132
           READ(502,*.END=400)
133 350
1.34
           N=N+1
135
           GO TO 350
136 400
           REWIND 502
137 0
138 0
139 0
        READ DATA FROM FUEL FILE, DATA INCLUDES:
140 C
141 0
           A(1,1) = ATMOSPHERIC PRESSURE
142 0
           A(1.2) = AMBIENT AIR TEMP, OF TEST(F)
143 C
           A(1,3) = WATER VAPOR PRESSURE AT A(1,2)
144 C
           A(I,4) = RELATIVE HUMIDITY
145 C
           A(1,5) = BRAKE HORSEPOWER
           A(1,6) = BRAKE SPECIFIC FUEL CONSUMPTION
146 C
147 0
           A(1,7) - BRAKE SPECIFIC VOLUME CONSUMPTION
148 C
149 C
150
           DO 16 I=1.N
           READ(502,+)(A(I.K),K=1,L)
151
152 0
153 C
154 0
        CALCULATE CORRECTION FACTOR(CF): AND CORRECT
155 C
        ENGINE PERFORMANCE DATA.
156 C
157 C
158
           R=A(1,4)/100.
159
           PV=R+A(1,3)
160
           BDT=A(1.1)-PV
161
           CF=(BD/BDT)+(((A(I,2)+AB)/F)++,7)
162 0
163 C
164 C
        OUTPUT CORRECTED VARIABLES
165 0
166 C
           B(I,1) = CORRECTED BHP
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167 0
          B(1,2) - CORRECTED BSFC
168 C
          B(1,3) = CORRECTED BSVC
          B(1,4) = CORRECTED SPECIFIC RANGE
169 C
170 C
          B(1,5) = CORRECTED % THERMAL EFFICIENCY
171 C
172 0
173
          B(I,1)=A(I,5)+CF
174
          FUG=A(1,5)/B(1,1)
175
          B(I,2)=A(I,6)+FUG
176
          B(1,3)=A(1,7)+FUG
          B(I,4)=1.2B(I,3)
177
          B(1,5)=100,+(1,2(BTLB+B(1,2)/2544,433))
178
179
          WRITE(LP, 15)IFNO, (B(I,M), M=1, J)
180 15
          FORNAT(24X,13,2X,1+1,2X,F6+2,2X,1+1,2X,F6+3,2X,1+1,2X,F6+4,
         12X, '+', 2X, F6.4, 2X, '+', 2X, F4.1)
191
182 16
          CONTINUE
183 0
184 C
185
          X=FLOAT(N)
          WRITE(LP, 17)
186
187 17
          FURMATC1X,/2/39X, 95% CONFIDENCE INTERVALS(12),37X,280(**)),20
         137X, 1+ LOW + NEAN + HIGH + DEVX+1,2,37X,28(1+1))
188
189 C
190 C
191 C
       CALCULATE MEANS, 95% CONFIDENCE INTERVALS. AND DEVIATION
192 C
       OF 95% CONFIDENCE INTERVALS AS A PERCENTAGE OF THE MEAN.
193 C
194 C
195
          DO 21 II=1,J
196
           J1 = II + J
197
           JK=II+2+J
198
          K2=II+3+J
199
           J2=II+4+J
200
          I2=II+5+J
201
          DO 18 KK=1,NCJ
202 18
          C(KK)=0.0
          DO 19 I=1,N
203
          C(II)=C(II)+B(I,II)
204
205
          C(J1)=C(J1)+B(I,II)++2,
206 19
          CONTINUE
207
          D(II)=C(II)/X
          SDEV=ABS(((C(J))-(C(II)++2,2X))/(X-1.)))++.5
208
209
          C( JK )=SDEV/( X++ . 5 )
210
          C(K2)=D(11)-2,+C(JK)
211
          C(J2)=D(II)+2,+C(JK)
212
          C(12)=((C(J2)-C(K2))/D(11))+100.
213
          IF(II, NE, 4)G0 T0 100
214 C
215 C
       OPEN SPECIFIC RANGE FILE.
216 C
217 C
       CHECK IF SPECIFIC RANGE FOR FUEL BLEND
218 C
       IS CONTAINED IN FILE.
219
   С
       WRITE TO SPECIFIC RANGE FILE IF NOT.
220 C
221 C
```

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222
           OPEN(505,FILE=NSRG)
223
           NF=0
224 97
           READ( $05, +, END=98)
           NF=NF+1
225
226
           GO TO 97
227 98
           REWIND 505
228
           DO 225 J=1, NF
           READ( 505, 60 )LFND, DEE, MAST
229
330
           IFCIEND EQ.LEND GO TO 100
231 225
           CONTINUE
232
           WRITE(505,60)IFNO,D(II),LAST
           FORMAT(1X, 13, 2X, F5, 2, 2X, 25A2)
233 60
234 100
           WRITE(LP,20)C(K2),D(II),C(J2),C(J2)
235 20
           FORMAT(37X,F6.3,1X,'+',F5.3,1X,'+',F5.3,1X,'+',F5.3,1X,'+',1X,F3.1)
236
           CLOSE ( 505 >
237 21
           CONTINUE
238 C
239 C
       DETERMINE IF SPECIFIC RANGE OUTPUT IS DESIRED
240 C
241 C
242 C
243
           WRITE(LU, 35)
244 35
           FORMATCIN, SUPPRESS SPECIFIC RANGE OUTPUT ? Y OR NO
245
           READ (LU, 36) NERF
           FORNAT(1A1)
246 36
247
           IFCHERF.NE. THY. AND. NERF. NE. THN JG0 TO 29
243
           IFCNERF.EQ.1HY)GO TO 25
           WRITE(LU.71)
249
           FORMATCIN, TYPE LU FOR HARD COPY OF SPECIFIC RANGE DATA--- ()
250 71
251
           READ(LU, +)L2
252
           OPEN(505,FILE=HSRG)
253
           ICHR=0
254 40
           READ(505, +, END=50)
255
           ICHR#ICHR+1
256
           GO TO 40
257 50
           REWIND 505
259
           IF. IFNO.EQ. NBF. AND.LFN.E0.1HY GO TO 75
259
           WRITE(LZ, 45)
          FORMATCIH1,2.4X,75(101),2,4X,101,1 (FUEL NO.1,4X, SPECIFIC RANGE),
13X,1% FROM BASELINE1,5X,1FUEL DESC (PTION1,7%,101,2,4X,75(101))
260 45
261
565 C
263 C
264 0
       OUTPUT SPECIFIC RANGE DATA AND % DEVIATION FROM BASELINE
265 0
       FOR A PARTICULAR FUEL BLEND.
266 0
267 0
269 75
           00 70 I=1,ICHR
269
           READ(505,60)LFN0, SRAN, LAST
270
           IF(LFN0.EQ.NBF)GG TO 80
271 70
           CONTINUE
272
           GO TO 120
273 30
           SPRN=SRAN
274
           REWIND 505
275
           IF(LFH,EQ.1HY)GO TO 175
276
           DO 110 1=1, ICHR
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277 READ(505, 60 MEND, SRAH, LAST 278 IF(LEND.EQ.NBF)G0 TU 110 279 DELT=(SRAN-SPRN)/SPRN+100. 230 WRITE(LZ, 150)LFN0, SRAN, DELT, LAST FORMATC10X, 13, 10X, F5.2, 10X, F5.1, 10X, 25A2> 201 150 282 110 CONTINUE 283 GO TO 140 284 175 DO 180 1-1, ICHR 285 READ(505.60)LFHO, SRAN, LAST IF(IFN0.EQ.NBF)G0 T0 140 286 IFELFNO.EQ.NOF DED TO 180 287 568 IFCIEND.EQ.LEND.GO TO 190 CONTINUE 289 180 CLOSE(505) 290 291 GO TO 23 DELT=(SRAN-SPRN)/SPRN+100. 292 190 WRITEKLZ, 150) JEND, SRAN, DELT, LAST 293 294 C 205 C 296 0 CLOSE ALL FILES, EXIT PROGRAM 237 0 298 0 299 GO TO 140 301 120 WRITE(LU, 130) 381 WRITE(L2,130) 302 130 FORMATCIN, 'NO BASELINES FOR COMPARISONS () 303 140 CL03E(505) IFKLP NE 6000 TO 25 304 305 WRITE(LU,22) FORMATCIX, 22, 18, PROGRAM COMPLETE, DATA OUTPUTS ON LINE PRINTER -306 22 307 GO TO 25 308 23 WRITE(LU,24)IFN0 309 24 FORMATC1X, 222, FUEL ELEND NO. ... (X, I3, 1X, THAS NOT BEEN TESTED +- ... 310 25 WRITE(LU, 115) FORMATCIN, "EXAMINE ANOTHER FUEL POLY OR NOT 311 115 312 READ(LU.116)NUT 313 116 FORMATCIAL> IFCHUT.NE. 1HY, AND, NUT. NE. 1HN 200 TO 29 314 315 IF(NUT.EQ.1HY)GO TO 450 316 29 CLOSE(542) 317 30 CLOSE(5 > 318 IF(LFN.EQ.1HY)G0 TO 26 319 CLOSE(503) 320 26 CLOSE(504) 321 END

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FTN4% COMPTLER: HP92834 REV.2030 (800821)

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+* PO WARNINGS ** NO ERRORS ** PROGRAM: 5525 COMMON: (HONE)

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AFUND T=00004 IS ON CR00037 USING 00002 BLKS R=0000

0001	FTN4,	L
0002		SUBROUTINE FFNO(1,NIT,B,ND,+)
0003		INTEGER ND(25)
0004		H=0
0005	1	READ(NIT,+,END=2)
0006		N=N+1
00ú7		GO TO 1
0000	2	REWIND NIT
0009		DO 10 IF=1,N
0010		READ(NIT,5)L,B,ND
0011	5	FORMAT(13,2X,F5.0,2X,25A2)
0012		IF(I.EQ.L)GO TO 20
0013	10	CONTINUE
0014		GO TO 30
0015	20	CONTINUE
0016		REWIND NIT
0017		RETURN
0018	30	CONTINUE
0019		REWIND NIT
0020		RETURN 1
0021		END

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APPENDIX F BASELINE NO. 1

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10000000000000000000000000000000000000	3.8	₹.	D.9	13.5	-10.0	ú	0.0	-11.6	-2.8	6.2	۱
		14.84	16.16	16.27	13.30	14.81	14.78	13.06	14.36	15.70	14.70
84444444444444 FUEL NO.	16	113	115	+	123	118	112	1 02	200	111	120

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: CAT 1-H AL-10115-F

********			*****	*******	*******
FUEL	BHP	BSFC	BSVC	SPECIFIC	THERMAL
	•			RANCE	
NO.		LB/HP-HR	GAL/HP-HR	HP-HR/G	AL EFF.
	*******	*******	********	********	} ****** ***
100 *	5.26 *	. 467	• .0660	+ 15,160	+ 29.2
100 +	5.22 +	. 474	• .0670	+ 14.933	+ 28.8
100 +	5.21 +		. 0681	+ 14.682	+ 28.3
100 +	5.21 +		+ . 0691	+ 14.682	+ 28.3
100 +	5.24 +		+ .0649	+ 15.413	* 29.7
100 -	5.24 +		+ .0659	+ 15,180	+ 29.2
100 +	5,15 +		+ .0673	+ 14.865	+ 28.9
100 +	5.13 *	. 497	+ .0693	+ 14,434	* 27.4
100 +	5.11 +	. 490	• ,0693	+ 14.440	• 27.9
100 +	5.06 +	. 497	• . 07 03	+ 14.232	* 27.4
100 +	5,15 +	. 472	+ ,0667	+ 15.002	+ 28.9
100 +	5,18 +	. 468	* . 0656	+ 15.232	• 29.2
100 +	5,15 +	. 491	+ .0682	+ 14.671	+ 28.3
100 +	5.14 +	. 479	+ .0671	+ 14.909	* 28.5
100 +	5.02 +	. 494	+ . 0686	+ 14.574	
100 +	5.04 +	. 496	+ . 0686	+ 14.574	
100 +	5.07 +	. 479	+ .0675	+ 14.818	* 28.5
100 +	5,04 +	. 490	+ .0675	+ 14.818	* 29.4
100 +	5.13 +	. 466	+ .0659	+ 15.178	+ 29.2
100 +	5.13 +	. 473	* .0669	+ 14.948	* 28.8
100 +	5.11 +	, 461	+ .0652	+ 15.342	+ 29.5
100 +	5.11 4	. 469	• .0662	+ 15,106	# 29.1
100 +	5,10 +	. 469	* .0663	+ 15.095	* 29.1
100 +	5,10 4	, 469	÷ .0663	+ 15.085	+ 29.1
100 +	5,13 +	. 487	+ .0689	+ 14.512	+ 28.0
100 +	5,13 4	. 487	+ . 0689	+ 14.512	# 28.0
100 +	5.15 4	, 493	* . 0697	+ 14.347	+ 27.7
100 +	5,15 4	, 496	+ . 0697	+ 14.558	* 28.1
100 *	5,18 4	, 476	+ ,0673	+ 14.967	+ 28.6
100 +	5.10 4	• .493	• .0683	* 14.649	• 29.2
100 +	5.15 4	.495	* .0687	+ 14.556	• 28.1
100 +	5.15		 .0677 	+ 14.773	+ 28.5
100 +	5.20	.480	+ .0690	+ 14.703	+ 28.4
100 +	5,30	• .480	+ .0690	+ 14.703	• 28.4
100 #	5.20 *	• .498	* .0690	+ 14.490	+ 27.9
100 +	V / E V	• , 488	+ .0690	+ 14,490	+ 27.9
100 +		• . 492	+ .0683	+ 14.652	• 29.3
100 +	5,15	• . 485	• .0687	+ 14.557	* 28 ,1

95% CONFIDENCE INTERVALS

5,130 +5,151 +5,171 + .8 .476 + .479 + .482 + 1.3 .067 + .068 + .068 + 1.3 14.688 +14.78 +14.88 + 1.3 26.291 +28.47 +28.66 + 1.3

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 84% AL-10115-F 10% DEIONIZED H20 6% EA78

FUEL		BHP		BSFC		BSVC		SPECIFIC Range	THERMAL
NO.				LB/HP-HR		GAL/HP-HR	2	HP-HR/GAL	EFF
****	***	******		*******	• •	*******	۰	******	******
102	٠	5.15	۰	. 560	۰	. 0769	۲	13.026 4	20.5
102		5.15		. 563		. 0778	٠	12.861 4	28.3
102	٠	5.15		. 560	٠	. 0767		13.030 .	28.4
102	*	5.15		. 560	+	. 0769	*	13.026 4	28.5
102		5.15		. 56 1	٠	. 0768	*	13.026 +	28.4
102		5.15		. 561	٠	. 0768	٠	13.026 +	28.4
102	#	5,18	٠	. 556		. 0763		13.114 .	28.7
102		5.18	•	. 555		. 0763	*	13.114 +	28.7
102		5.18		. 554	٠	. 0763	•	13.114 +	28.8
102	•	5.10	•	. 553	٠	. 0763		13.114 +	28.9
102	•	5.18		. 555	٠	. 0763		13.110 +	28.7
102	**	5.19		. 535	*	. 0763	*	13.110 +	

95% CONFIDENCE INTERVALS + LOW + MEAN + HIGH + DEV%+ 5.155 +5.165 +5.175 + .4 .556 + .558 + .560 + .7 .076 + .077 + .077 + .7 13.013 +13.06 +13.10 + .7 28.480 +28.58 +28.68 + .7

PROJECT NO. 05-5957-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 99% JP-10 AL-9220-F 1% ANYL NITRATE

*****	****	******	***	*******	•	******		********	*******
FUEL		внр		BSFC		BSYC		SPECIFIC Range	THERMAL
NO.				LB/HP-HR		GAL/HP-HR		HP-HR/GAL	EFF.
*****	****	******	***	********	ala a	*******	*	********	*****
111	•	5,09	٠	. 501	٠	. 0644	٠	15.525	27.5
111	•	5.13		, 495	٠	. 0631		15.836	27.9
111	٠	5,13		, 496		. 0631	aşta	15.837	27.9
111	٠	5.11		. 499		. 0642	٠	15.586	
111	٠	5,13				. 0631	•	15.837	
111	٠	5,13		. 502	٠			15.586	27.5

95% CONFIDENCE INTERVALS + LOW + MERN + HIGH + DEVX+ 5,102 +5,116 +5,129 + .5 .496 + .498 + .501 + .9 .063 + .064 + .064 + 1.6 15,579 +15,70 +15.82 + 1.6 27,565 +27.69 +27.82 + .9

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 90.99% DF-2 AL-9979-F 3.01% PHENANTHRENE

*****	***	******	*	*******	1 1 1	********	* 4	*******	F 4	******
FUEL		BHP		BSFC		BSVC	S	PECIFIC		THERMAL
								RANGE		
NO.				LB/HP-HR		GAL/HP-HR		HP-HR/G/	AL	EFF.
*****	***	*******	*	*******	•	********	+ 4	*******	Þ¢	******
113	٠	5.27	۰	. 492	۰	. 0679	•	14.710		28.2
113		5.25		. 492	٠	. 0679	•	14.718	٠	28.2
113		5.26		.403	٠	. 0670	•	14.915		28.7
113		5.26	۰	. 484	٠	. 0672	٠	14.092		28.6
113	٠	5.24		. 487		. 0672	•	14.892		28.5
113		5.25	٠	.481	٠	. 0672	٠	14.892		28.8

95% COHFIDENCE INTERVALS + LOU + MEAN + HIGH + DEV%+ 5.245 +5.255 +5.265 + .4 .483 + .486 + .490 + 1.5 .067 + .067 + .068 + 1.0 14.762 +14.84 +14.91 + 1.0 28.285 +28.59 +28.71 + 1.5

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 80.58% DF-2 AL-9979-F 19.42% NAPTHALENE

FUEL		BHP		BHP BSFC			BSVC	•	BPECIFIC RANGE	THERMAL		
NO.				LB/HP-HR		GAL/HP-HR		HP-HR/GA	L	EFF		
112	****	5.10	***	.511	*** *	********** . 0703	1823 184	14,219	中川 山	27.4		
112	•	5.11	-	. 489		. 0671	÷		-	28.6		
112	•	5.18		. 403		. 0660			•	29.0		
112		5/21		. 535		. 0736		13.578	÷	26.2		
112		5.21		, 479	۰	. 0656	۰	15,250	+	29.2		
112	•	5.22		. 487		. 0667	۰	14,997	۰	28.9		
112	٠	5.09		.490		. 0675	۰	14.823	•	28.6		
112	•	5.09		, 482	+	. 0664	٠	15.051		29.1		
112		5.10	٠	. 483		. 0662	٠	15,101	•	29.0		

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 50% CAT 1-H AL-10115-F 50% RJ5G AL-7159-F

+++++	***	******		*******	in sile s	*****	a in the second	******	•	******
FUEL		BHP		ÐSFC		BSVC	ł	SPECIFIC RANGE		THERNAL
NO.				LB/HP-HR		GAL/HP-HR	2	HP-HR/GA	L	EFF.
*****	***	******		********	h aife a	*******		******	٠	******
114		5.26		. 491	۰	. 06 09	٠	16.433	٠	28.2
114		5.27	٠	. 485		. 0598	•	16.713	•	28.5
114		5.27		, 485	٠	. 0598	٠	16,713	٠	28.5
114	•	5.27		. 484	٠	. 0598	+	16.713	•	29.6
114	٠	5.29		. 477	٠	. 0587	•	17.028		29.0
144	٠	5.28	•	. 477	*	. 0587	٠	17.020	٠	29.0

95% CONFIDENCE INTERVALS + LOW + NEAN + MIGH + DEVX+ 5.269 +5.278 +5.286 + .3 .479 + .483 + .487 + 1.8 .059 + .060 + .060 + 2.2 16.586 +16.77 +16.96 + 2.2 28.369 +28.63 +28.89 + 1.8

PROJECT NO. 05-5057-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL OLEND: 75% TETRALIN 24% AL-10115-F 1% ANYL NITRATE

*****	***	******	**	*******	ih s	******		*******		*****
FUEL		BHP		BSFC		BSYC		SPECIFIC		THERNAL
NO.				LB/HP-HR		GAL/HP-HR		RANGE HP-HR/GA	ι	EFF.
*****	***	******	-	********		*******	۰	*****	*	******
115	-	5.23		. 486		. 0618	•	16.173		29.4
115		5.23		. 495		.0618	ışı.	16,173	*	29.5
115		5.23		. 484	•	.0619	+	16,163	*	29.5
115	٠	5.23		. 494	•	. 0619	•	16.163		29.5
115	٠	5,23		, 484			*		•	29.5
115		5.22			+	. 0620	•	16.136		29.5

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 72% JP-10 25% NAPTHALENE 3% ANYL NITRATE

*****	in alle alle a	******	**	******	n alfa a	******	+	*****	*******
FUEL		BHP		BSFC		BSVČ		SPECIFIC Range	THERMAL
NO.				LB/HP-HR		GAL/HP-HR	2	HP-HR/CA	
******	1. ap ap a	*******	***	********	n aite a	****	- HA	****	*****
116		5.12	240	.514	۰	.0640	-	15.617	+ 28.2
116	+	5.12	*	. 494		.0620	*	16.129	+ 29.4
116	•	5,12	-	. 528		. 0661	*	15.136	+ 27.5
116		5.12		. 525	*	. 0661	+	15.136	+ 27.6
116		5.12		. 528	*	. 0661	-	15.136	* 27.5
116	•	5.11	•	. 532	۰	. 0671	۰	14.894	+ 27.3

95% CONFIDENCE INTERVALS + LOW * MEAN * HIGH * DEVX* 5.115 *5.115 *5.115 * 0.0 .509 * .520 * .532 * 4.4 .064 * .065 * .067 * 4.7 14.972 *15.34 *15.71 * 4.8 27.292 *27.93 *28.57 * 4.6

PROJECT NO. 05-5957-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 63% FRF(#102) 37% RJSG AL-7159-F

******		******	**	*******		********		*********	*****
FUEL		BHP		BSFC		BSAC	1	SPECIFIC 1 Range	HERNAL
NO.				L8/HP-HR		GAL/HP-HR	2	HP-HR/GAL	EFF .
******	***	******	**	*******	*	********	, alju	*********	*****
118	•	5.29		. 542	٠	. 0691	٠	14.674 +	27.8
119		5.29		. 537		. 0671	٠	14.893 +	28.1
119	•	5.29		. 542		. 0681	•	14.674 +	27.8
118		5.28		. 538	٠	. 0672	*	14.983 +	29.1
118		5.28		. 533	٠	. 0672		14.893 +	28.4
118	•	5.28		. 538	۰	. 0672		14.893 #	28.1

95% CONFIDENCE INTERVALS + LOW + MEAN + HIGH + DEV%+ 5.287 >5.287 + 5.287 + 0.0 .535 + .538 + .541 + 1.1 .067 + .068 + .068 + 1.2 14.726 +14.81 +14.90 + 1.2 27.899 +28.05 +28.21 + 1.1

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: SOX FRF(#102) 50% JP-10

*****	***	******	**	********) () (*******	**	*******	*****
FUEL		BHP		BSFC		BSVC	9	PECIFIC RANGE	THERMAL
NO.				LB/HP-HR		GAL/HP-HR		HP-HR/GA	EFF.
*****	***	*******	-	********	•	*******	++	*******	*****
120		5.30		.513	٠	. 0680	٠	14.703	28.8
120	٠	5.20		. 523	٠	. 0680	•	14.703	28.2
120		5.30		.513	٠	. 0680	*	14.703	28.8
120	•	5.30		.513	۰	. 0680	•	14.703	28.8
120		5.30		.513		. 0680		14.703	28.8
120	٠	5.20	٠	.523		. 0690	۰		28.2

95% CONFIDENCE INTERVALS • LOU * NEAN * HIGH * DEVX* 5.223 *5.266 *5.308 * 1.6 .512 * .516 * .521 * 1.6 .068 * .069 * .068 * .1 14.703 *14.73 *14.70 * 0.0 28.374 *29.61 *28.84 * 1.6

PROJECT NO. 05-5057-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 85.6% FRF-A 14.4% NAPTHALENE

******	***	******	***	*******	\$ 1	*****	. 18	*****	******
FUEL		BHP		BSFC		BSYC	1	SPECIFIC Range	THERMAL
NO.				L8/HP-HR		GAL/HP-HR	2	HP-HR/GA	L EFF.
*****	***	******	***	*******	4	*******	÷	********	*******
123		5.15		. 560		. 0757	٠	13.202	. 28.4
123		5.15		. 558		. 0747	*	13.380	+ 28.5
123	٠	5.17	٠	, 558		. 0755	٠	13,249	• 28.5
123		5.15		. 554		. 0747		13,380	+ 28.6
123		5,15		. 554	•	. 0747		13,380	+ 29.6
123		5.15		. 560		. 0757		13,202	+ 28.4

95% CONFIDENCE INTERVALS + LOW + MEAN + HIGH + DEV% 5.146 +5.152 +5.157 + .2 .555 + .557 + .559 + .7 .075 + .075 + .076 + 1.1 13.225 +13.30 +13.37 + 1.1 28.407 +28.50 +28.60 + .7

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: JET-A AL-10112-F

FUEL		BHP		BSFC		BSVC		SPECIFIC T RANGE	HERMAL
NO.				LB/HP-HR		GAL/HP-H	₹	HP-HR/GAL	EFF
****	****	******	a de ales	******	h # 1	******	**	*****	*****
200	+	5.21		. 465		. 0699	-	14.307 +	29.0
200	*	5.21	*	. 466		. 0699	*	14.307 +	28.9
200	*	5.21		.467	*	. 0699	*	14.312 +	28.9
200		5.21		. 465		. 0689			29.0
200	alfit.	5.21		. 464		. 0689	Ma	14.524 *	29.1
200		5.21		. 464	+	. 0689	*	14.524 *	29.1
200		5.17	*	. 467		. 0694	*	14.406 +	28.9
200	-	5.15	*	. 468		. 0697	*	14.357 *	28.8
200		5.17	. de	. 467		. 0694		14.406 *	28.9
200		5.15	*	. 468	-	. 0697	aft	14.357 *	28.8
200	+	5.15	*	. 468	s‡t	.0697	Ma	14.357 *	28.8
200		5.15		. 477		. 0717		13.953 +	28.2

95% CONFIDENCE INTERVALS

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APPENDIX G • BASELINE NO. 2 ÷.

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	FUEL DESCRIPTION .	***************************************	79% DF-2 6% EA-78 5% RJ-5 10% WATER	15% NAPHTHALENE 85% CAT-1H	80% TETRALEN 19% SAT-1H 1% ANYL NITRATE	81% DF-2 3.5% EA-78 10% RJ-5 5% WATER	00% 45 BURNER 01L 20% JP-10	79% 46 BURNER OIL 20% JP-10 1% ANYL NITRATE	89% 46 BURNER OIL 10% CAT-1H 1% AMYL NITRATE		R.J-6M	97% TETRALIN AL-9762-A 3% ANYL NITRATE	84% AL-10115-F 10% DEIONIZED M20 6% EA78	84% AL-10115-F 6% EA-78 5% RJ-5 5% H20(DI)	80% JP-10 20% MAPTHALENE	86.5% CAT-1H 3.5% EA-8 5% RJ5 5% (DI)H20
*************	GE X FROM BASELINE	************	-10.8	m.	4.2	14.2	6	4.8	5.4	2.3	11.9	7.7	-12.5	-2.1	7.1	2.6
	SPECIFIC RANGE	~~~~~	13.67	15.36	15.97	14.67	15.41	16.05	16.15	15.67	17.15	16.50	13.41	15.00	16.41	15.72
	EUEL NO.		144	176	175	150	167	169	170	168	182	117	102	156	164	149

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: CAT 1-H AL-10115-F

			******	*****	● ski ski ski ski ski ski
**********	8HP	BSFC	BSVC SP	ECIFIC T	HERMAL
FUEL	BUL	•		RANGE	
NO.	La	/HP-H R G	AL/HP-HR	HP-HR/GAL	EFF.
********	******	********	. 0634 +	15.779 +	30.4
100 •	5,30 +	.449 + .452 +	.0640 +	15.633 +	30.1
100 +	5.30 +	.452 +	.0639 +	15.657 +	30.1
100 +	5.31 +	.451 +	. 0639 +	15.657 +	30.2
100 +	5,31 + 5,31 +	.452 +	. 0639 *	15.657 *	30.1
100 + 100 +	5,30 +	.450 +	. 0637 +	15.706 +	30.3
100 + 100 +	5.22 +	.469 +	.0661 +	15.125 +	29.1
100 +	5.21 +	,465 +	.0657 +	15.217 +	29.3
100 +	5.21 +	.469 +	. 0662 🔹	15.102 +	29.1
100 +	5.21 +	.470 +	.0663 +	15.080 +	29.0
100 +	5.25 +	,439 +	.0620 *	16.134 +	31.1
100 +	5.22 *	.477 +	.0674 *	14.829 +	28.6
100 +	5.22 +	.463 +	. 0654 *	15.293 +	29.4 29.9
100 +	5.22 +	,455 +	. 0644 +	15,522 *	29.9
100 🕶	5.21 +	.456 +	.0645 +	15.740 +	30.3
100 🕶	5.24 +	.450 +	.0635 +	15,619 +	30,1
100 +	5.25 +	,453 + .477 +	,0640 * ,0673 *	14.862 +	28,6
100 +	5.22 +	1 4 1 4	.0659 *	15.178 +	29.2
100 +	5.21 *	.467 *	,0652 *	15.326 +	29.5
100 *	5.25 *	.456 +	. 8645 +	15,515 *	29.9
100 + 100 +	5.26 + 5.33 +	.448 *	.0633 +	15.796 #	30.4
	5.32 +	.448 +	.0633 +	15.796 *	30.4
100 + 100 +	5.23 +	.463 +	.0654 *	15.291 *	29.5
100 +	5.23 +	.459 +	,0647 *	15.446 *	
100 +	5.25 *	.462 *	,0652 *	15,334 +	
100 +	5.24 *	.464 +	, 0655 *	15,264 *	
100 +	5.22 +	.480 +	.0678 +	14.750 +	
100 +	5.23 +	,471 +	,0666 *	15.015 *	
100 #	5.20 *	.472 +	, 0669 *	14.980 +	
100 +	5.21 +	. 471 +	, 0667 +	15.002 4	
100 +	5,17 *	.475 +	.0671 +	15,035	
100 +	5.17 +	.470 +	.0665 + .0650 +	15.383	
100 +	5.13 +	.460 + .457 +	.0645 +	15,499	
100 +	5.19 +	.458 *	.0647 +		29.8
100 +	5.21 +	.457 +	.0646 +		29.8
	5.23 * 5.23 *	.457 +	,0645 +	14144	29.8
100 + 100 +	5,17 +	.469 +	.0662 *	15,104	• 29.0
100 +	5,14 *	.459 +	,0648 +		+ 29.7
100 +	5,17 *	,467 +	,0659 +		• 29.2
100 +	5,19 +	.457 +	.0646 *		• 29.9
100 +	5.12 *	.475 +			* 28.7
100 +	5,12 *	,469 *			* 29,1 * 29,1
100 +	5.10 +	.469 *			+ 29.1 + 29.2
100 +	5.11 +	. 466 +			* 28.7
100 *	5.12 *	. 475 +			• 29.1
100 +	5.12 +	.469 *			+ 29.1
100 +	5.10 +	.469 *			_ ,

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95% CONFIDENCE INTERVALS

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• LOV • MEAN • HIGH • DEVX• 5.194 •5.211 •5.220 • .7 .460 • .462 • .465 • 1.1 .065 • .065 • .066 • 1.1 15.229 •15.32 •15.40 • 1.1 29.326 •29.49 •29.66 • 1.1

PROJECT NO. 05-5957-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 04% AL-10115-F 10% DEIGNIZED HRO 6% EA70

******				*******			•4	*******	*******
FUEL		BHP		BSFC		8 87C	6	PECIFIC RANGE	THERMAL
NO.						GAL/HP-HR		-	-
1 02	•••	5,19				. 0732			
102	•	5.19	٠	. 540	۰	. 0751	٠	13.321	• 29.1
102	۰	5.23	٠	. 551	٠	. 0756	٠	13.233	• 29.9

95% CONFIDENCE INTERVALS • LOW • MEAN + NIGH • DEVX+ 5.179 •5.206 •5.233 • 1.0 .534 • .544 • .554 • 3.9 .073 • .075 • .076 • 3.9 13.143 •13.41 •13.67 • 3.9 28.739 •29.31 •29.07 • 3.9

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL PLEND: 97% TETRALIN AL-9762-A 3% ANYL HITRATE

*****	****	******	***	********	****	*****	**	****	****	******
FUEL		BHP		BSFC	89	VC	S	PECIF		THERMAL
								RANG	E	
NO.				L8/HP-HR	GAL	/HP-HF	1	HP-H	IR/GAL	L EFF.
	****	******	**	*********	****	*****		*****	****	*******
117	٠	5.26	٠	. 494 🔸		06 08	٠	16.4	46	+ 29.7
117	•	5.26		.492 +		06 85		16.5	27	• 29.8
117		5.26		.497 +		0599	٠	16.6	92	+ 30.1
117	٠	5.25		. 496 +		0610	٠	16.3	192	+ 29.6
117		5.24		.497 +		0611	٠	16.3	165	• 29.5
117		5.26		. 491 +		06 04	٠	16.5	554	♦ 29.9

95% CONFIDENCE INTERVALS + LOU + MEAN + HIGH + DEV%+ 5.244 +5.250 +5.257 + .3 .489 + .492 + .495 + 1.2 .060 + .061 + .061 + 1.2 16.397 +16.50 +16.59 + 1.2 29.569 +29.75 +29.93 + 1.2

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

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FUEL BLEND: 79% DF-2 6% EA-78 5% RJ-5 10% WATER

						****	*	******	a nije s	******
FUEL	****	8HP		BSFC	• •	BSVC	Ş	SPECIFIC RANGE		THERMAL
NO.				LB/HP-HR		GAL/HP-HR		HP-HR/GA		. EFF ******
*****	****	******** 5.26	**) *		∓ - •	. 0732		13.670	٠	29.0
144	Ξ.	5.23	*		+	. 0736	1	13.578	۰	28.8
144		5.25		.542	•	. 0730	*	13,708	*	29.1 29.1
144	٠	5.24		.541		, 0729	٠	13,726	*	27.1

95% CONFIDENCE INTERVALS

* LOW + MEAN * HIGH * DEV%* 5.234 *5.247 *5.260 * .5 .540 * .543 * .546 * 1.0 .073 * .073 * .974 * 1.0 13.605 *13.67 *13.74 * 1.0 20.861 *29.00 *29.14 * 1.0

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 86.5% CAT-1H 3.5% EA-8 5% RJ5 5% (DI >H20

				والمراجعة والمراجعة والمراجعة والمراجعة	اد مد	*****	***	******	aje s	******
++++++ FUEL	in sila sila s	8HP	4 4 4 4	8SFC	-	BSVC	SI	PECIFIC RANGE		THERMAL
NO.				LB/HP-HR		GAL/HP-HR	de de :	HP-HR/GA		EFF.
	***	******	ala ala a	*********		***	· • • • •		<u>.</u>	31.1
149		5.24	+	.471	10	. 0643	*	15.554	۰	
•••	-		-	• • • •		. 0635		15,749	*	31.5
149	*	5.24	•				-	15,724	*	31.5
149		5.23		. 466		, 0636	+			• • •
	•	5.24		. 466	٠	, 0635	- 19	15,749	*	31.5
149			•	• • •		. 0633	*	15.799	-	31.7
149		5.24	- 4	. 464	*		-		•	
149	*	5.24		, 465	*	. 0634	-	15,774		31.0

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 81% DF-2 3.5% EA-78 10% RJ-5 5% WATER

*****	***	*****	**	*******	k alfe a	********	1	********	ф. ф . н	*****
FUEL		BHP		BSFC		BSAC	٩	SPECIFIC RANGE		THERMAL
NO.				LB/HP-HR		GAL/HP-HR	2	HP-HR/G	AL	EFF .
*****	***	****	- 44 - 44 - 2	********	la alfa a	********	i aļt i	******	***	******
150		5.20	*	. 506	٠	. 0690	#	14.702	۰	29.0
150		5.20	+	.507	s#F	. 0682	*	14.658		29.0
150		5,19		. 598	۰	, 0683	۰	14.637	٠	28.9

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 84% AL-10115-F 6% EA-78 5% RJ-5 5% H20(DI)

*****	***	******	aju ajt a	****	0 10		**	******	-	******
FUEL		өнр		BSFC	9	SYC		SPECIFIC RANGE	•	THERMAL
NO.				L8/HP-HR	GA	L/HP-H	R	HP-HR/GA	L	EFF.
*****	****		***	****	***	*****	• +	******	# 1	******
156		5,31		. 489 🖪		.0670	*	14.931		30.4
156		5.30		.489 4		. 0668		14.975		30.5
156	+	5,32		.495 4		. 0664		15.064		30.7
156		5,31		. 486 4		. 0666		15.019		30.6
156	*	5.31		. 486 4	ja i	. 0666		15.019	٠	30.6
156	*	5,30		.487 4	Þ	. 0667		14,997	۰	30.5

95% CONFIDENCE INTERVALS + LOU + MEAN + HIGH + DEV%+ 5.306 +5.311 +5.317 + .2 .486 + .497 + .488 + .5 .067 + .067 + .067 + .5 14.964 +15.00 +15.04 + .5 30.472 +30.55 +30.62 + .5

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 80% JP-10 20% NAPTHALENE

*****	***	******	**	********	n alju a	*******	-	******	140 X4	*****
FUEL		BHP		BSFC		BSVC		SPECIFIC RANGE		THERNAL
NO.				LB/HP-HR		GAL/HP-HR	2	HP-HR/G	AL	EFF.
*****	***	*******	**	******	in alfan a	******	l WE	******	**	******
164	-	5.20	۰	. 485	۰	.0610	*	16.390	-	29.4
164		5.21		. 482		.0607	ų	16.471		29.6
164	•	5.19		. 492		. 06 07	٠	16.471		29.6
164		3.19		. 482.		.0607	+	16.471		29.6
164		5.17	*	.478	*	.0602	+	16.609	-	29.9
164	-	5.16		. 496	۰	. 0624	+	16.020		28.8

95% CONFIDENCE INTERVALS

+ LOU + MEAN + HIGH + DEV%+ 5.175 +5.190 +5.205 + .6 .480 + .484 + .489 + 2.0 .060 + .061 + .062 + 2.0 16.241 +16.41 +16.57 + 2.0 29.200 +29.49 +29.78 + 1.9

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 80% #5 BURNER OIL 20% JP-10

*****	***	******	**	*******	a alfa a	*****	. ada	Na nia sia sia sia sia sia sia sia sia	للمراجع المراجع المراجع والمراجع
FUEL		BHP		BSFC		BSVC		SPECIFIC	THERMAL
NO.				L8/HP-HR		GAL/HP-HR	2	RANGE HP~HR/GA	L EFF.
*****	****	******	**	********	u de s	*******		****	ala ala ale ale ale ale ale ale
167	+	5,16		. 503	٠	. 0654			+ 28.4
167	+	5.23		. 500		.0650	*		■ 28.5
167	•	5.23		, 499					+ 28.6
167		5.20		, 498					* 28,6
167		5,21	r#	. 497	*				+ 28.7
167	ada -	5,19	÷		+				• 28,7

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

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FUEL BLEND: 90% #5 BURNER OIL 10% CAT-1H

*****	****	******	.	******	e alfe a	*******	He	*******	k +	*****
FUEL		BHP		BSFC		BSYC		SPECIFIC RANGE		THERNAL
NO .				L87HP-HR		GAL/HP-HR	2	HP-HR/G	٩L	. EFF.
*****	****	******	***	********	in adja a	*~*****	1 1 1	*******	k 4	******
168	*	5,17	*	. 495		. 0645	-	15.515		28.4
168	*	5,18		. 491	-	. 0638	÷	15.662	4	28.7
168	*	5.18	*	. 488	alp.	. 0635	ale .	15.736	4	28.9
169	*	5,18	*	, 488	٠	. 0634	*	15.761		28,9
168	+	5,18	*	. 492	*	. 0641	*	15.613		28.6
169	•	5,17		.487		. 0634	÷	15.761	•	28.9

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 79% #6 BURNER OIL 20% JP-10 1% AMYL NITRATE

****	***	****	a alla alla a	********	k 🕸 X	*******	-	*****	1 A	******
FUEL		BHP		BSFC		BSVC		SPECIFIC		THERMAL
								RANGE		
NO.				LB/HP-HR		GAL/HP~HR	2	HP-HR/GA	IL	EFF.
*****	**	i de de de de de de de	-	*******	je sje s	*******	r sje	*****) 1	******
169		5.23	-	. 504		. 0623		16,047		29.0
169		5.24		.507		. 0627		15.945	#	28.8
169		5.23		.504	-	,0623	1	16.047	sķe	29.0
169		5.23		. 504		. 0622	*	16,073		29.0
169	*	5.23		.502	*	. 0621	*	16,098		29.1
169		5.23		.504		. 0622		16.073		29.0

95% CONFIDENCE INTERVALS + LOW + MEAN + HIGH + DEV%+ 5.225 +5.227 +5.229 + .1 .502 + .504 + .505 + .5 .062 + .062 + .062 + .5 16.004 +16.05 +16.09 + .5 28.928 +29.00 +29.08 + .5

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 89% 46 BURHER OIL 10% CAT-1H 1% ANYL HITRATE

*****	****	*****	1 1 1 1 1 1	******	k a ta a	******	*	*****	k 14	a an an an an an an an an
FUEL		BHP		BSFC		BSVC	1	SPECIFIC RANGE		THERMAL
NO.				LB/HP-HR		GAL/HP-HR		HP-HR/GF	ìL	EFF.
*****	****		a de adra	*****	ik alju a	*******	4	*****	1 - A	*****
170	•	5.23		. 494		. 0615	٠	16.268		29.3
170	•	5.24		. 497	٠	. 0618	*	16.190		29.1
170	-	5.23		. 502		. 0624	+	16.035		28.8
170		5,25		. 497		. 0618		16.190		29.1
170	-	5.24		.498			•	16.164		29.1
170		5.22	*	.501				16.035	*	28,9

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 80% TETRALEN 19% CAT-1H 1% ANYL NITRATE

			•	1420	-		-	13,973	-	4 0.0	
175	•	5.22		.500		. 0631		15,845		29.7	
175		5.23	*					15.794			
175	-	5.21	ağı:					15,819			
175	aļa 🛛	5.20						16.230			
175	+	5.21						16.230			

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CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL BLEND: 15% NAPHTHALENE 85% CAT-1H

******	1 1 1 1 1	*****	***	*******	l ağı a	******	-	*****	*****
FUEL		BHP		BSFC		BSYC		SPECIFIC	THERMAL
								RANGE	
NO.				L8/HP-HR		GAL/HP-HR	2	HP-HR/GAI	
*****	***	******	**	********	t sije s	******	: 3 4 0	******	******
176		5.17		, 479	1	. 0656	#	15.243	+ 28.8
176	*	5,18	*	. 476	-	. 0652	-	15.338	* 28.9
176	*	5,17	+	. 473	*	. 0648	4	15.433	* 29.1
176		5.17		. 475	۰	. 0651		15.362 .	• 29.0
176	ф.	5,18		. 474	-	.0650	٠	15.396	• 29.1
176		5.17	+	. 475	۰	. 0650	*	15.385 -	¥ 29.0

95% CONFIDENCE INTERVALS

* LOU * NEAN * HIGH * DEV2* 5.172 *5.175 *5.179 * .1 .474 * .476 * .477 * .7 .065 * .065 * .065 * .7 15.305 *15.36 *15.41 * .7 28.871 *28.98 *29.08 * .7

PROJECT NO. 05-5857-130

CORRECTED PETTER ENGINE DATA FOR FUEL BLENDS

FUEL GLEND: RJ-6M

*****	***	*****	a alian a dia s	*****		****	aja aj	***		an an an an an an an an an
FUEL		BHP		BSFC		BSYC	ę	PECIFIC RANGE		THERMAL
NO.				LB/HP-HR		GAL/HP-HR		HP-HR/GA	L	EFF.
*****	***	*******	-	******		*******	nie si	*****		a an an an an an an an an
192		5.11		. 499		. 0584		17.111		29.0
182		5.15		. 485	-	.0581		17,203	*	29.2
192		5,13		. 487	+		•	17,141	÷	29.1
162	+	5.14		. 486			*	17,171	÷.	29.1
182		5,15	۰	. 487	*			17.143	•	29.0

95% CONFIDENCE INTERVALS

G-11

NO. 1 CARBON BLACK (RAVEN 1170) NO. 2 CARBON BLACK (SL-90) NO. 3 CARBON BLACK (MOGUL L)

NO.1

Properties	of	Raven	1170 ^(a)
(Cities	s S	ervice	Co.)
(Cities	J S	ervice	

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- Talet	160
Blackness Index	
	24
Particle Diameter, m (arithmetric mean)	
Surface Area, sq. meters per gram (by nitrogen adsorption (B.E.T.)	102
Ofl (Venuto Method)	7.2
Absorption, Oll (Veram Fluid Paste, cc/gram gal./100 lb	86.
Absorption, 011	.89
Absorption, 011 Stiff Paste Endpoint, cc/gram gal./100 lb	8.3
Absorption, BDP, cc/100 grams	58
Tinting Strength Index	102
Tinting Science	6.0
Adsorption Index	7.0
DN .	99.0
Fixed Carbon, % Volatile matter, %	1.0
Apparent Density, 1bs/cu.ft.	17
Powder	30
	13,875
Beads Net Hc(b) Btu/1b Ash (b) wt%	0.5



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NO. 2

PROPERTIES (a) OF SL-90 CARBON BLACK

(ASHLAND CHEMICAL CO.)

0.3
N11
0.0111
0.39
93
24.8
57.5
42.8
32.3
6.4
8.2
33
22
4
39.0

(a) As supplied by manufacturer

(b) D Numbers in parentheses are ASTM Test Methods

NO. 3

PROPERTIES^(a) OF MOGUL CARBON BLACK (CABOT CORPORATION)

Nigrometer Index	83
Surface Area (N.S.A.), sg. meters per gram	138
Particle Size, milli microns	24
Oil Absorption (DBS), cc/100 grams	
Fluffy	60
Pellets	55
Tinting Strength Index	112
Volatile Content, %	5.0
Fixed Carbon, X	95.0
PH	3.4
Toulene extract, %	0.10
Electrical Resistivity (dry)	H
Density, lbs/cu.ft.	
Pluffy	15
Pellets	33
Net Hc ^(b) , Btu/1b	13,240

(a) As supplied by manufacturer

(b) Experimentally determined at AFLRL

DEPARTMENT OF DEFENSE

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DEFENSE DOCUMENTATION CTR CAMERON STATION ALEXANDRIA VA 22314	12
DEPT OF DEFENSE ATTN: DASD(MRAL)-LM(MR DYCKMAN) WASHINGTON DC 20301	1
COMMANDER DEPENSE LOGISTICS AGY ATTN DLA-SME (MRS P MCLAIN) CAMERON STATION ALEXANDRIA VA 22314	1
COMMANDER DEFENSE FUEL SUPPLY CTR ATTN: DFSC-T (MR. MARTIN) CAMERON STA ALEXANDRIA VA 22314	1
COMMANDER DEFENSE GENERAL SUPPLY CTR ATTN: DGSC-SSA RICHMOND VA 23297	1
DOD OFC OF SEC OF DEF ATTN USD (R&E) WASHINGTON, DC 20301	1
DOD ATTN OASD (MRA&L)-TD PENTAGON, 3C841 WASHINGTON DC 20301	1
DEFENSE ADVANCED RES PROJ AGENCY DEFENSE SCIENCES OFC 1400 WILSON BLVD ARLINGTON VA 22209	I
DEPARTMENT OF THE ARMY	
HQ, DEPT OF ARMY ATTN: DALO-TSE (COL ARNAUD) DALO-AV DALO-SMZ-E DAMA-CSS-P (DR BRYANT) DAMA-ARZ (DR CHURCH) DAMA-SHZ	1 1 1 1 1

WASHINGTON DC 20310

CDR U.S. ARMY MOBILITY EQUIPMENT R&D COMMAND 10 Attn: DRDME-GL FORT BELVOIR VA 22060 CDR US ARMY MATERIEL DEVELGREADINESS COMMAND ATTN: DRCLD (MR BENDER) 1 DRCDMR (MR GREINER) 1 DRCDMD-ST (DR HALBY) 1 DRCQA-E 1 DRCDE-SS 1 DRCIS- (LTC CROW) 1 5001 EISENHOWER AVE ALEXANDRIA VA 22333 CDR US ARMY TANK-AUTOMOTIVE CMD ATTN DRSTA-NW (TWVMO) DRSTA-RG (MR HAMPARIAN) DRSTA-NS (DR PETRICK) 1 DRSTA-G 1 DRSTA-M 1 DRSTA-GBP (MR MCCARTNEY) 1 WARREN MI 48090 DIRECTOR US ARMY MATERIEL SYSTEMS ANALYSIS AGENCY ATTN DRXSY-CM DRXSY-S DRXSY-L 1 ABERDEEN PROVING GROUND MD 21005 DIRECTOR APPLIED TECHNOLOGY LAB U.S. ARMY R&T LAB (AVRADCOM) ATTN DAVDL-ATL-ATP (MR MORROW) 1 DAVDL-ATL-ASV (MR CARPER) 1 FORT EUSTIS VA 23604 HQ, 172D INFANTRY BRIGADE (ALASKA) ATTN AF2T-DI-L 1 1 APZT-DI-M DIRECTORATE OF INDUSTRIAL OPERATIONS

FT RICHARDSON AK 99505

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CDR US ARMY GENERAL MATERIAL & PETROLEUM ACTIVITY ATTN STSGP-F (MR SPRIGGS) STSGP-PE (MR MCKNIGHT), BLDG 85-3 STSGP (COL CLIFTON) NEW CUMBERLAND ARMY DEPOT NEW CUMBERLAND PA 17070 CDR US ARMY MATERIEL ARMAMENT READINESS CMD ATTN DRSAR-LEM (MR MENKE) ROCK ISLAND ARSENAL IL 61299 CDR US ARMY COLD REGION TEST CENTER ATTN STECR-TA APO SEATTLE 98733 HQ, DEPT. OF ARMY ATTN: DAEN-RDZ-B WASHINGTON, DC 20310 CDR US ARMY RES & STDZN GROUP (EUROPE) ATTN DRXSN-UK-RA BOX 65 FPO NEW YORK C9510 HQ. US ARMY AVIATION RED CMD ATTN DRDAV-GT (MR R LEWIS) 1 DRDAV-D (MR CRAWFORD) 1 DRDAV-N (MR BORGMAN) 1 DRDAV-E (MR LONG) 1 4300 GOODFELLOW BLVD ST LOUIS MO 63120 CDR US ARMY PORCES COMMAND ATTN AFLC-REG AFLG-POP FORT MCPHERSON GA 30330 CDP US ARMY ABERDEEN PROVING GROUND ATTN: STEAP-MT-U (MR DEAVER) 1 ABERDEEN PROVING GROUND MD 21005 CDR US ARMY YUMA PROVING GROUND ATTN STEYP-MT (MR DOEBBLER) YUMA AZ 85364

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MICHIGAN ARMY MISSILE PLANT OFC OF PROJ MGR, ABRAMS TANK SYS ATTN DRCPM-GCM-S 1 WARREN MI 48090 MICHIGAN ARMY MISSILE PLANT PROG MGR, FIGHTING VEHICLE SYS ATTN DRCPM-FVS-SE 1 WARREN MI 48090 PROJ MGR, M60 TANK DEVELOPMENT USMC-LNO, MAJ. VARELLA US ARMY TANK-AUTOMOTIVE CMD (TACOM) WARREN MI 48090 PROG MGR, M113/M113A1 FAMILY OF VEHICLES ATTN DRCPM-M113 1 WARREN MI 48090 PROJ MGR, MOBILE ELECTRIC POWER ATTN DRCPM-MEP-TM 1 7500 BACKLICK ROAD SPRINGFIELD VA 22150 OFC OF PROJ MGR, IMPROVED TOW VEHICLE US ARMY TANK-AUTOMOTIVE R&D CMD ATTN DRCPM-ITV-T 1 WARREN MI 48090 CDR US ARMY EUROPE & SEVENTH ARMY ATTN AEAGC-FMD 1 APO NY 09403 PROJ MGR, PATRIOT PROJ OFC ATTN DRCPM-MD-T-G 1 US ARMY DARCOM **REDSTONE ARSENAL AL 35809** CDR THEATER ARMY MATERIAL MONT · CENTER (200TH) DIRECTORATE FOR PETROL MGMT ATTN AEAGD-MM-PT-Q (MR PINZOLA) 1 ZWEIBRUCKEN APO NY 09052 CDR US ARMY RESEARCH OFC ATTN DRXRO-ZC 1 DRXRO-EG (DR SINGLETON) 1 DRXRO-CB (DR GHIRARDELLI) 1 P O BOX 12211 RSCH TRIANGLE PARK NC 27709

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DIR US ARMY R&T LAB (AVRADCON) ATTN DAVDL-AS (MR D WILSTED) 1 NASA/AMES RSCH CTR MAIL STP 207-5 MOFFIT FIELD CA 94035 CDR TOBYHANNA ARMY DEPOT AITN SDSTO-TP-S 1 TOBYHANNA PA 18466 DIR US ARMY MATERIALS & MECHANICS RSCH CTR ATTN DRXMR-EM 1 URXMR-R 1 DRXMR-T I WATERTOWN MA 02172 CDR US ARMY DEPOT SYSTEMS CMD ATTN DRSDS 1 CHAMBERSBURG PA 17201 CDR US ARMY WATERVLIET ARSENAL TN SARWY-RDD 1 WATERVLIET NY 12189 CDR US ARMY LEA ATTN DALO-LEP 1 NEW CUMBERLAND ARMY DEPOT NEW CUMBERLAND PA 17070 CDR US ARMY GENERAL MATERIAL & PETROLEUM ACTIVITY ATTN STSGP-PW (MR PRICE) 1 SHARPE ARMY DEPOT LATHROP CA 95330 CC . US SAY PU MA SCIENCE & TECH 17 275 ATL CLUT MIL 1 PEDT AT LLD" CDR DARCOM MATERIEL READINESS SUPPORT ACTIVITY (MRSA) ATTN DRXMD-MD 1 LEXINGTON KY 40511

HQ, US ARMY TEE COMMAND ATTN DRSTE-TO-O 1 ABERDEEN PROVING GROUND, MD 21005 HQ, US ARMY ARMAMENT RED CHD ATTN DRDAR-LC 1 DRDAR-SC 1 DRDAR-AC 1 DRDAR-OA 1 DOVER NJ 07801 HQ, US ARMY TROOP SUPPORT & AVIATION MATERIAL READINESS COMMAND ATTN DRSTS-MBG (2) DRCPO-PDE (LTC FOSTER) 1 4300 GOODFELLOW BLVD ST LOUIS NO 63120 DEPARTMENT OF THE ARMY CONSTRUCTION ENG RSCH LAB ATTN CERL-EM 1 CERL-ZT 1 CERL-EH Ł P O BOX 4005 CHAMPAIGN IL 61820 DIR US ARMY ARMAMENT R&D CMD BALLISTIC RESEARCH LAB ATTN DRDAR-BLV 1 DRDAR-BLP 1 ABERDEEN PROVING GROUND, MD 21005 HQ US ARMY TRAINING & DOCTRINE CMD ATTN ATDO-5 (COL MILLS) 1 FORT MONROE VA 23651 DIRECTOR US ARMY RSCH & TECH LAB (AVRADCOM) PROPULSION LABORATORY ATTN DAVDL-PL-D (MR ACURIO) 1 21000 BROOKPARK ROAD CLEVELAND OH 44135 CDR US ARMY NATICK RES & DEV CMD ATTN DRDNA-YEP (DR KAPLAN) 1 NATICK MA 01760 CDR US ARMY TRANSPORTATION SCHOOL ATTN ATSP-CD-MS 1 FORT EUSTIS VA 23604

12/81 AFLRL No. 147 Page 3 of 6 CDR US ARMY QUARTERMASTER SCHOOL ATTN ATSH-CD (COL VULPE) 1 ATSM-CDM 1 ATSM-TNG-PT 1 FORT LEE VA 23801 HQ. US ARMY ARMOR CENTER ATTN ATZK-CD-SB 1 FORT KNOX KY 40121 CDR US ARMY LOGISTICS CTR ATTN ATCL-MS (MR A MARSHALL) 1 FORT LEE VA 23801 CDR US ARMY FIELD ARTILLERY SCHOOL ATTN ATSF-CD 1 FORT SILL OK 73503 CDR US ARMY ORDNANCE CTR & SCHOOL. ATTN ATSL-CTD-MS ABERDEEN PROVING GROUND MD 21005 CDR US ARMY ENGINEER SCHOOL ATTN ATSE-CDM 1 FORT BELVOIR VA 22060 CDR US ARMY INFANTRY SCHOOL ATTN ATSH-CD-MS-M 1 FORT BENNING GA 31905 CDR US ARMY AVIATION BOARD ATTN ATZQ-OT-C 1 ATZQ-OT-A 1 FORT RUCKER AL 36362 CDR US ARMY MISSILE CMD ATTN DRSMI-O 1 DRSMI-RK 1 DRSM1-D 1 **REDSTONE ARSENAL, AL 35809** CHIEF US ARMY LOGISTIC ASSISTANCE OFFICE (TSARCOM) ATTN STSPS-OE (LTC BRYANDS, SSTR) 1 P.O. BOX 2221 APO NY 09403 12/81

MAJOR L E GUNNIN, SSTR US ARMY LOGISTIC ASSISTANCE OFFICE LAO-K (TSARCOM) APO SAN FRANCISCO 96202 CRD US ARMY AVIATION CTR ATTN ATZQ-D 1 FORT RUCKER AL 36362 PROJ MCR M60 TANK DEVELOP. ATTN DRCPM-M60-E (MR WESAK) 1 WARREN MI 48090 CDR US ARMY INFANTRY BOARD ATTN ATZB-1B-PR-T 1 FORT BENNING, GA 31905 CDR US ARMY FIELD ARTILLERY BOARD ATTN ATZR-BDPR 1 FORT SILL OK 73503 CDR US ARMY ARMOR & ENGINEER BOARD 1 ATTN ATZK-AE-PD ATZK-AE-CV 1 FORT KNOX, KY 40121 t CDR US ARMY CHENICAL SCHOOL 1 ATTN ATZN-CM-CS FORT MCCLELIAN, AL 36205 DZPARTMENT OF THE NAVY CDR NAVAL AIR PROPULSION CENTER ATTN PE-71 (MR WACNER) 1 PE-72 (MR D'ORAZIO) 1 P O BOX 7176 **TRENTON NJ 06828** CDR NAVAL SEA SYSTEMS CMD CODE 05D4 (MR R LAYNE) 1 WASHINGTON DC 20362 CDR DAVID TAYLOR NAVAL SHIP R&D CTR CODE 2830 (MR G BOSMAJIAN) 1 CODE 2831 1 ANNAPOLIS MD 21402

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JOINT OIL ANALYSIS PROGRAM -1 TECHNICAL SUPPORT CTR **BLDG 780** NAVAL AIR STATION PENSACOLA FL 32508 DEPARTMENT OF THE NAVY HQ, US MARINE CORPS 1 ATTN LPP (MAJ SANDBERG) LMM (MAJ STROCK) 1 WASHINGTON DC 20380 CDR NAVAL AIR SYSTEMS CHD ATTN CODE 5304C1 (MR WEINBURG) 1 CODE 53645 (MR MEARNS) 1 WASHINGTON DC 20361 CDR NAVAL AIR DEVELOPMENT CTR ATTN CODE 60612 (MR L STALLINGS) 1 WARMINSTER PA 18974 CDR NAVAL RESEARCH LABORATORY ATTN CODE 6170 (MR H RAVNER) 1 1 CODE 6180 1 CODE 6110 (DR HARVEY) WASHINGTON DC 20375 CDR NAVAL PACILITIES ENGR CTR ATTN CODE 1202B (MR R BURRIS) 1 CODE 120B (MR BUSCHELMAN) 1 200 STOVALL ST ALEXANDRIA VA 22322 CHIEF OF NAVAL RESEARCH 1 ATTN CODE 473 ARLINGTON VA 22217 CDR NAVAL AIR ENGR CENTER ATTN CODE 92727 LAKEHURST NJ 08733 CDR, NAVAL MATERIAL COMMAND ATTN MAT-083 (DR A ROBERTS) MAT-OSE (MR ZIEM) CP6, RM 606 WASHINGTON DC 20360

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CDR NAVY PETROLEUM OFC 1 ATTN CODE 40 CAMERON STATION ALEXANDRIA VA 22314 CDR MARINE CORPS LOGISTICS SUPPORT BASE ATLANTIC 1 ATTN CODE P841 ALBANY GA 31704 DEPARTMENT OF THE AIR FORCE HQ, USAF 1 ATTN LEYSF (MAJ LENZ) WASHINGTON DC 20330 HO AIR FORCE SYSTEMS CMD ATTN AFSC/DLF (LTC RADLOF) 1 ANDREWS AFB MD 20334 CDR US AIR PORCE WRIGHT AERONAUTICAL LAB ATTN AFWAL/POSF (MR CHURCHILL) 1 1 AFWAL/POSL (MR JONES) 1 AFWAL/MLSE (MR MORRIS) 1 AFWAL-MLBT WRIGHT-PATTERSON AFB OH 45433 CDR USAF SAN ANTONIO AIR LOGISTICS CTR ATTN SAALC/SFQ (MR MAKRIS) 1 ١ SAALC/MMPRR KELLY AIR FORCE BASE, TX 78241 CDR USAF WARNER ROBINS AIR LOGISTIC CTR ATTN WR-ALC/MMIRAB-1 (MR GRAHAM) 1 ROBINS AFB GA 31098

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OTHER GOVERNMENT AGENCIES

US DEPARTMENT OF TRANSPORTATION ATTN AIRCRAFT DESIGN CRITERIA BRANCH 2 FEDERAL AVIATION ADMIN 2100 2ND ST SW WASHINGTON DC 20590

US DEPARTMENT OF ENERGY DIV OF TRANS ENERGY CONSERV ALTERNATIVE FUBLS UTILIZATION BRANCH 20 MASSACHUSETTS AVENUE WASHINGTON DL 20545

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DIRECTOR

NATL MAINTENANCE TECH SUPPORT CTR US POSTAL SERVICE Norman ok 73069 US DEPARTMENT OF ENERGY BARTLESVILLE ENERGY RSCH CTR DIV OF PROCESSING & THERMO RES DIV OF UTILIZATION RES BOX 1398 BARTLESVILLE OK 74003

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SCI & TECH INFO FACILITY ATTN NASA REP (SAK/DL) 1 P O BOX 8757 BALTIMORE/WASH INT AIRPORT MD 21240

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