



ELECTRODE BOUNDARY LAYERS IN DENSE, DIFFUSE PLASMAS

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This technical report has been reviewed and is approved for publication.

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FOREWORD

This final report was prepared by Oscar Biblarz and S. T. van Brocklin of the Department of Aeronautics of the Naval Postgraduate School, Monterey, CA 93940 for AFWAL under MIPR 79-00618, Project 2308, task \$503, Dr. Alan Garscadden was the Government Project Monitor. This report has been identified within the Naval Postgraduate School by the number NPS67-80-011 and was submitted to the Air Force on July of 1980.

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SECTION I

INTRODUCTION

Electrode boundary regions play an important role in high energy density discharges presently used for electrical lasers. Discharge media of particular interest are molecular gases in glow-discharge generated plasmas as exemplified by the ${\rm CO_2}$ laser. A quantitative description of the electrode regions is required in order to establish energy loadings and geometric scaling in such discharges for various practical devices presently under development.

The electrode regions correspond to those portions of the plasma that are affected by the presence of the electrodes. 1 Under certain conditions, the relevant characteristic lengths of the problem are independent of and smaller than typical interelectrode dimensions. The variables, therefore, display a "boundary layer" behavior. This behavior is already well understood for any fluid dynamic lengths that may be of importance. 2 In both stationary and moving plasmas of moderately high density, the electrode regions are of the boundary layer type and the electrode contributions may be described with the continuum or classical fluid dynamical equations. In addition to diffusion³, these equations must describe the electrical conduction term, Poisson's equation, and suitable energy relations. To the extent that the presence of nonequilibrium effects may be well represented, the continuum approach may be both simpler than present kinetic theory

approaches⁴, ⁵ and accurate. Any classical approach brings together a vast body of literature unmatched by kinetic theory, even if the latter is more appropriate for non-equilibrium situations. The method in our work corresponds to a hybrid approach to the problem.

Electrode voltage drops are significant in glow discharges because in addition to the necessary allowance required in the external power source, "one can never impose a uniform electric field on a plasma by means of collecting electrodes". 6 Thus, in the ionizer/sustainer discharge mode of pumping a laser, the proper E/N value is obtained after the electrode regions have been accounted for. Moreover, the values of E/N are higher in the electrode regions than in the plasma core (i.e., in the positive column or undisturbed plasma), and the discharge stability to glow collapse may be governed by what E/N values can be tolerated in the electrode regions without exceeding some ionization rate critical value which will lead to instabilities and arcing. 8 Another phenomenon that affects both plasma homogeneity and stability is the presence of constriction or spots at the electrodes9 surfaces. In the so-called vacuum arc, 10,11 anode spots appear under certain operating conditions and are of some practical importance; anode spots have also been observed 12 in high density plasmas. Since intense vacuum arcs produce a dense metal vapor at the surface, they too are described by the continuum equations and ultimately it should be possible to reconcile results from the arc literature to our work.

The intense cathode spot is the conventional mode of operation of many arcs; however, we shall be only concerned with describing the "cold cathode" because of its relative simplicity and its closer kinship with anode behavior.

Thus, the imposition of a homogeneous, precisely defined value of E/N in an E-beam generated plasma is not without challenge. The E-beam generated ionization is usually less homogeneous than the sustainer field distribution so that E/N and $n_{\rm e}$ may vary in space and time throughout the discharge. For CW devices, we must add fluctuations in density due to turbulence and to heating, and in pulsed devices fluctuations in density due to shock waves. Fortunately, in the $\rm H_{\rm e}\textsc{-N_2}\textsc{-CO_2}$ laser, the optimum E/N for laser efficiency is rather broad.

1. Approach

Our description of the anode and the (cold) cathode is based on the existence of both a continuum and local thermodynamic equilibrium (LTE)¹³. As such, except for photons, particles are locally at equilibrium and the Boltzmann relation between energy states may be used. Moreover, we sometimes assume that both electrons and heavy particles have Maxwellian velocity distributions but at different temperatures. Whenever electron collisions are the primary source of transitions, then the electron temperature appears in the corresponding Boltzmann relation. Thus, the non-equilibrium situation represented is one where the electrons have a Maxwellian distribution at a corresponding temperature greater than the translational or vibrational temperatures of the lasing molecules. The term "two-temperature plasma" is often used to describe the above situation¹³.

At this stage, there are a number of simplifications incorporated in our description. First, the effects of a magnetic field are neglected since no external magnetic fields are present and since in the unconstricted mode the internal or self-generated magnetic fields is expected to have no effect on the discharge. Second, we consider the overall gas temperature to remain constant; this is an adequate assumption for convectively cooled, high-volume discharges. In such a thermally stable system, however, transitions from a glow to arc filaments cannot be represented. Third and last, the molecular gas described is pure nitrogen rather than the required laser mixture, and the source of ionization through an E-beam or other source is external to the sustainer power; the only function of the electrodes is that the imposing an E/N to pump the laser medium.

The equations that describe the electrode voltage drops comprise species continuity, the electron energy equation, and Poisson's equation^{15,16}. The small electron mass yields important consequences, two of which are the ability to neglect convection in the electron continuity equation and the strong dependence of the electron temperature on E/N. In this report, we decouple the presence of the E-beam by assuming that the sustainer operates only during the afterglow region.

In glow discharges, the electron energy distribution is known to be non-Maxwelliar. 17-19 and suitable account must be made of this important fact. In order to allow for non-Maxwellian distributions and still retain the use of the concept of temperature,

empirical information is used in the representation of the coefficients 20 ; thus, the electron temperature, the ionization and recombination coefficients, the diffusion coefficients, etc., are given as parametrized functions of E/N from experimental measurements.

In the usual boundary layer problem in fluid mechanics a two-dimensional flat plate is assumed². It is shown herein that similarly a two-dimensional Cartesian description is the minimum suitable description for our equations in the limit of low currents. We further argue that convection can only be significant in the plasma portion of the ambipolar regions.

Note that compared to the sheath, the ambipolar regions contribute only a minor fraction of the voltage drop. The sheath and ambipolar regions, however, are important in questions of stability of the discharge and must be included in the problem formation.

We model a quiescent plasma in the immediate neighbourhood of the electrode surfaces. In an attempt to describe CW devices, we have included in this report a discussion of possible effects of convection. In actuality, no calculations of such effects are performed. Now in pulsed lasers, where pulse lengths of the order of 50 µsec are of interest, convection will play no direct role. Of considerable importance, however, is the fact that both the sheath and ambipolar regions can establish themselves within such short times; this possibility is a direct result of the boundary layer nature of the problem.

2. Parameters of Discharge Gas

Throughout this work we deal exclusively with nitrogen discharges under the following coefficients:

Gas Density: 1 Amagat or $3x10^{25}$ m⁻³

Gas Type: molecular (nitrogen)

Gas Temperature: 273 OK

Electric Field: 1.5 to $15x10^5$ V/m (E/N = 0.5 to $5x10^{-20}$

 Vm^2)

Electron Density: 10^{17} to 10^{19} m⁻³

Electrode Voltag Drop: 35V

Since cross-section data for electron in nitrogen discharges are readily available 18, the electron temperature and diffusion coefficient are calculable. Also, n*/N where n* is a stationary state (see Section II-2) and N the gas number density can be computed. These parameters are shown in Figure 1 as a function of E/N. The ions are assumed to follow the gas temperature. An electrode voltage drop of about 35 volts may be assumed since it represents the effective ionization energy for N2 and, therefore, the anode fall. However, results shown in Sec. IV are presently restricted to 10 volts because of computational difficulties at the higher voltages.

The cathode is more complicated to describe because of the requirement of electron emission from the surface. The cathode fall in cold cathodes is of the order of 250 volts and in thermionic cathodes considerably less. A proper representation of the cathode will require a more efficient computational scheme with suitable boundary conditions.

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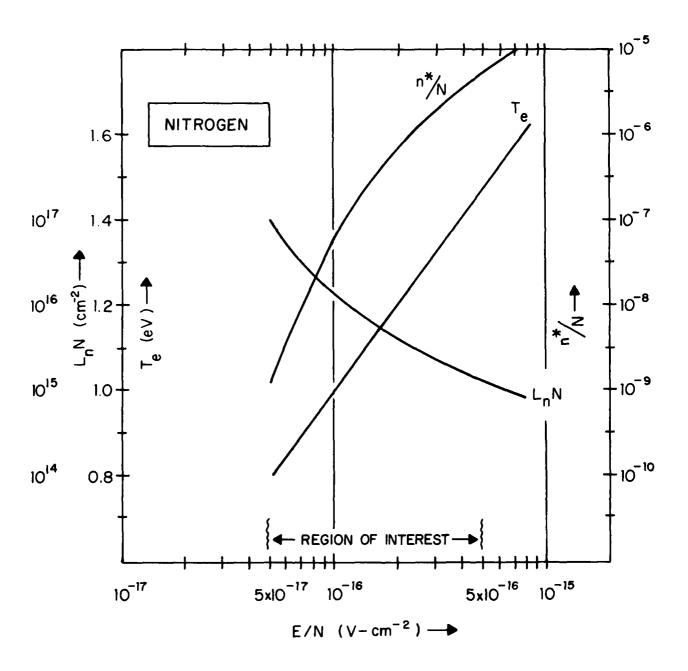


Figure 1: Nitrogen Parameters for E/N Range of Interest

SECTION II

PROBLEM FORMULATION

1. Equations

The species equations for singly ionized atoms without any magnetic fields may be written as are commonly in mass transfer analysis $^{1.5}$, $^{1.6}$, $^{2.1}$

Ions:
$$\rho \tilde{\mathbf{u}} \cdot \nabla C_{\mathbf{i}} - \nabla \cdot \left[\rho D_{\mathbf{i}} \left(-\frac{eC_{\mathbf{i}}\tilde{\mathbf{E}}}{kT_{\mathbf{o}}} + \frac{C_{\mathbf{i}}}{P_{\mathbf{i}}} \nabla p_{\mathbf{i}} \right) \right] = \dot{\mathbf{w}}_{\mathbf{i}}$$
 (1)

Electrons:
$$-\frac{m_e}{e} \cdot q - \nabla \cdot \left[\rho D_e \left(\frac{e C_e \overline{E}}{k T_e} + \frac{C_e}{p_e} \nabla p_c \right) \right] = \dot{w}_e (2)$$

where γ = overall gas density

u = overall gas velocity field

 $C_{i,e} = \text{species mass fraction } \Sigma_{j}C_{j} = 1 \quad (\rho C_{j} = m_{j}n_{j})$

 $\dot{\mathbf{w}}_{i,e}$ = net source term $\Sigma_{j}\dot{\mathbf{w}}_{j}$ = 0

 $D_{i,c} = diffusion coefficient$

 T_{o} = electron temperature

 T_{O} = overall gas temperature

 $p_{i,e}$ = species partial pressure $(p_j = n_j kT_j)$

e = charge of the electron

 m_{e} = mass of the electron

k = Boltzmann's constant

E = electric field

q = rate of thermalization of fast electrons

Futhermore,

$$\dot{w}_i = \dot{w}_e$$
 for single ionization in the gas (3)

and
$$E = -\nabla \phi$$
 where ϕ is the electric potential (4)

In our notation, n represents species number density and N the overall number density. Also, since $T_{\rm e}$ is a given function of E/N, there is no further need for an electron energy equation. The species mass fraction, C, is in itself dimensionless but it will prove useful to define a new ratio for this variable (see Section III). Equations 1 and 2 are written for steady flow.

The only equation needed to complete the set now is Poisson's equation,

$$\nabla^2 \Phi = -\frac{e\rho}{\varepsilon_0^{m_i}} (C_i - \frac{m_i}{m_e} C_e)$$
 (5)

where ε_0 = permittivity of free space.

From this set of equations, the sheath and ambipolar regions evolve in a self-consistent way, obviating the requirement to match boundary conditions between the regions.

2. Characteristic Lengths

It turns out that all the dependent variables in this problem, namely, C_i , C_e , and $\nabla \phi$, can be of the boundary-layer type. As such, there is an individual characteristic length over which the magnitude of these variables changes from their value at the electrode to their value in the undisturbed plasma. Of course, the fluid dynamic boundary

layer is established by non-electrical considerations and we shall assume that it ranges from one or a few millimeters to a few centimeters depending on the Peynolds number of the flow.

The concentration boundary layers are not easily surmised because of inflections at the sheath and because, as will be pointed out, their boundary layer nature depends strongly on the degree of reaction in the plasma, i.e., on $\dot{w}_{\rm p}$.

The extent of the sheath is one of the most important characteristic lengths in this work because it is within the sheath that most of the voltage drop is expected to occur (for a short discharge). Fortunately, the sheath length can be estimated rather easily from Poisson's or in this case Gauss' equation,

$$\nabla \cdot \bar{E} = \frac{e}{\varepsilon_{O}} (n_{i} - n_{e})$$
 (6)

where $\ \epsilon_{_{\hbox{\scriptsize O}}}$ is the permittivity of free space.

Let
$$\hat{E} = \frac{(E - E_S)^{\lambda}_S}{|\phi_S - \phi_{\infty}|} \text{ and } \hat{n}_{i,e} = \frac{n_{i,e}}{n_{\infty}}$$
 (7)

where $\lambda_{\rm S}$ = sheath characteristic length; the subscript "s" indicates the value at the electrode (anode or cathode) and " ∞ " the value at the undisturbed plasma. Equation 7 then becomes

$$\hat{\nabla} \cdot \hat{\mathbf{E}} = \begin{bmatrix} \frac{e n_{\infty} \lambda_{s}^{2}}{\epsilon_{o} |\phi_{s} - \phi_{\infty}|} \\ \hat{\mathbf{n}}_{i} - \hat{\mathbf{n}}_{e} \end{bmatrix} (\hat{\mathbf{n}}_{i} - \hat{\mathbf{n}}_{e})$$
 (8)

In the equation above, $\lambda_{\rm S}$ is used to non-dimensionalize the del operator in all directions. That is, independent of the dimensionality of the problem, the sheath is effective over the characteristic dimension of $\lambda_{\rm S}$. Now we can estimate the size of the sheath by setting the square brackets in Equation 8 to be of order one. Since all other terms have been made of order one,

$$\lambda_{s} = \left(\frac{\varepsilon_{o} |\phi_{s} - \phi_{\infty}|}{e n_{\infty}}\right)^{\frac{1}{2}} = \lambda_{D_{\infty}} \left(\frac{e |\phi_{s} - \phi_{\infty}|}{k T_{e_{\infty}}}\right)^{\frac{1}{2}}$$
(9)

where λ_{D_m} = the Debye length in the undisturbed plasma.

We now establish the specific values anticipated for $\lambda_{_{\mathbf{S}}}$. Take for example

$$n_{\infty} = 10^{17} \text{ to } 10^{19} \text{m}^{-3}$$

$$|\phi_{S} - \phi_{\infty}| = 35 \text{ volts}$$
 Then,
$$\lambda_{S} = 1.4 \times 10^{-4} \text{ to}_{1.4} \times 10^{-5} \text{m}$$

It is interesting to note that a change of $|\phi_{_{\rm S}} - \phi_{_{\infty}}|$ by a factor of 10 would change $\lambda_{_{\rm S}}$ by a factor of 3 so that we may consider the estimate reasonable for the above-quoted electron number densities in the undisturbed plasma. Now, if the fluid dynamic boundary layer for a density of one Amagat is at least an order of magnitude greater than the sheath, then we may conclude that convection will have a small if not

negligible effect in the sheath. Note that the sheath, being always adjacent to the surface, resides in the region where the velocity drops to zero; therefore, it is entirely appropriate to assume that convection plays no significant role in the sheath.

Next in importance is the length of the ambipolar region, i.e., the transition region between the sheath and the undisturbed plasma. This region is neutral and can perhaps span the boundary layer so that, in CW devices, convection would likely be present. We do, however, expect that convection may be neglected without affecting the resulting voltages appreciably. In Equation 1, for example, we know²³ that in the ambipolar region the last term of the left hand side is small compared to electric conduction; therefore, we need only compare convection with conduction, or the fluid velocity with the drift velocity. The transverse velocity component in the boundary layer problem is²

$$v \stackrel{\sim}{\sim} U_{\infty} R_{e}^{-\frac{1}{2}} \stackrel{\sim}{\sim} 0.1-lm/s$$

assuming the Reynolds number to be $~R_{\rm e}$ 2 10 4 - 10 6 and the free stream velocity to be $~U_{\infty}$ 2 10 - 100 m/s . Now the drift velocity may be approximated by

$$v_D \stackrel{\sim}{\sim} \mu_i E_\infty \stackrel{\sim}{\sim} 10 - 100 \text{ m/s}$$

for $E_{\infty} \stackrel{?}{\sim} 10^5$ - 10^6 V/m and $\mu_{\hat{i}} \stackrel{?}{\sim} 10^{-4}$ m²/s·volt. Clearly, in the presence of a sufficiently strong interelectrode field,

the contribution of ion convection due to a cross flow should be negligible. The geometrical orientation of the discharge with respect to the flow is an important factor here. 9

The concentration profiles behave in a more complicated manner than either the voltage or the electric field. While the magnitude of n_i and n_e will change appreciably within the sheath, fractional analysis is risky because inflections are present in the profiles. In this section and in Appendix A, we give a discussion pertaining to the stability of the ambipolar region and its boundary layer nature. Since L_n (see Equation A6 and Figure 1), the characteristic length for ambipolar diffusion, turns out to be comparable to the sheath length, ambipolar diffusion to the walls can establish itself during periods of the order of 50 μ sec.

In general, the ambipolar diffusion equation can be written as

$$\frac{\partial \mathbf{n}}{\partial t} - \mathbf{D_a} \nabla^2 \mathbf{n} = \dot{\mathbf{n}}_{\mathbf{e}} \tag{10}$$

where D_a = ambipolar diffusion coefficient If we may assume a form for the net production rate, \dot{n}_e , some important conclusions may be drawn about the stability of the ambipolar region. Let

$$\dot{n}_{\Theta} = v_{i} n - \alpha n^{2} + \Psi$$
 (11)

where

 v_i = ionization rate coefficient, sustainer

α = three-body neutral recombination rate coefficient

Ψ = ionization rate coefficient E beam

In a nitrogen discharge, no electron-neutral attachment is (Presumably, 24 attachment instabilities can be significant in lasers, but are not described here.)

If we look now at the steady form of the equation we have

$$\nabla^{2} n = -\frac{\alpha n}{D_{a}} (n*-n) - \Psi/D_{a}$$
 (12)
where
$$n* = \nu_{i}/\alpha \quad \text{or} \quad n*/M = \frac{\nu_{i}/N}{\alpha}$$

The parameter n* above is governed solely by the sustainer discharge, i.e., by E/N as shown in Figure 1. It is reasonable to assume that E/N is designed to be fairly homogeneous in the interelectrode space but that the E-beam produced charge concentration is not 14. Now at the outer edge of the ambipolar region (at the undisturbed plasma), $n(x,y) = n_m$, where n_m is assumed to be governed exclusively by the E-beam conditions. If we now assume that we operate in the afterglow region.

$$\nabla^2 n_{\infty} = \frac{\alpha n_{\infty}}{D_a} (n * - n_{\infty})$$
 (13)

From the properties of the Laplacian operator, we can infer the following:

- i) for $n^* > n_{\infty}$, $\nabla^2 n_{\infty} < 0$ and a steady solution is possible: n can reach a maximum.
- ii) for $n^* < n_{\infty}$, $\nabla^2 n > 0$ and the steady solution is not possible; n cannot be a maximum.

iii) for $n* = n_{\infty}$, n can neither be a maximum nor a minimum inside the domain and, indeed, it must be a constant.

The reason that the second case is impossible is that in the ambipolar region $\nabla^2 n < 0$ and n seeks a maximum toward the centerline. If, however, $n* < n_\infty$ then this trend must reverse itself somewhere and a minimum must exist within the domain. Such a situation corresponds to a physically unsteady condition and the full form of Equation 10 must be investigated.

The criterion for stability appears to be $\;n^*>n_{_{\!\infty}}$, with the equal sign as the marginally stable case. $\;n_{_{\!\infty}}\;$ is considered to be a steady distribution attributed solely to the E-beam; this simplification of the physics models the ionizer/sustainer as if the plasma is externally generated and a steady distribution ($n_{\!\!\!_{\!\infty}}$) is independent of E/N and reasonably constant for the pulse duration.

In pulsed lasers, diffusion may only be established within the boundary layers. This, however, is in itself significant because the electric field reaches a maximum within the electrode regions. But the problem cannot be simplified here and one must work with the entire set of equations.

The use of the ambipolar diffusion equation, i.e., Equation 10, is the traditional approach to the problem²⁵. This, however, rather complicates the calculation of the electric field as it does in the Cchottky solution.²⁵ This complication does not seem to appear when the full

formulation of the problem including the sheath, is utilized. In order to get some intuition into the nature of the solutions, it is worthwhile to attempt some sort of simplification. In Appendix A, the ion flux equation in the ambipolar region is investigated; it is assumed that this equation is sufficient to describe the concentration profile given an electric field and the form of \hat{n}_e . We further assume that a one-dimensional description is possible and, indeed, desirable since it yields a constant electric field in this region. Results indicate that for E/N values of interest the ambipolar charge concentration is indeed of the boundary layer type.

3. Dimensionality of the Problem

In the limit of low currents, the elevation of electron temperature may be neglected together with ionization from the sustainer discharge, and the quiescent plasma governing equations may be written in a simpler, equivalent form. We shall further examine the one dimensional form because, at first glance, the smallness of the sheath compared to a typical electrode dimension suggests that the one-dimensional Cartesian description might indeed be adequate. As before, y is the interelectrode coordinate.

$$j_{i} = \frac{e^{2}D_{i}}{kT_{O}} n_{i}E - eD_{i} \frac{dn_{i}}{dy}$$
 (16)

$$j_{e} = \frac{e^{2}D_{e}}{kT_{o}} n_{e}E + eD_{e}\frac{dn_{e}}{dy}$$
 (17)

$$\frac{dE}{dy} = \frac{e}{\varepsilon_0} (n_i - n_e)$$
 (18)

Now in one-dimensional flow with no ionization or recombination, j_i and j_e are individually constant throughout the interelectrode space.

Subtracting Equation 17 from 16 we obtain

$$K^{-} = \frac{\dot{J}_{i}}{eD_{i}} - \frac{\dot{J}_{e}}{eD_{e}} = \frac{eE}{kT_{o}} (n_{i} - n_{e}) - \frac{d}{dy} (n_{i} + n_{e})$$
 (19)

Now let us look at the sign of the terms in the above equation for three regions, namely. (a) in the anode sheath, (b) in the undisturbed plasma, and (c) in the cathode sheath.

Region	$\frac{eE}{kT_0}(n_i-n_e)$	$-\frac{\mathrm{d}}{\mathrm{dy}}(\mathrm{n_i} + \mathrm{n_e})$	к-
(a)	negative	negative	negative
(b)	0	0	0
(c)	positive	positive	positive

Clearly, if j_i , j_e , D_i , and D_e are constants, then Equation 19 is invalid. What is needed is for j_i and j_e to decrease from the anode to the plasma and to increase from the plasma to the cathode thereby producing current concentrations at the electrodes¹². It is therefore obvious that for the frozen-property flow of current a one-dimensional formulation is inadequate.

In order to see if ionization is a requisite for a onedimensional solution we use the conservation equations in the form of Equation 19 and we get

$$\frac{d}{dy} \left[\frac{j_i}{eD_i} - \frac{j_e}{eD_e} \right] = \hat{n}_e \left(\frac{1}{D_i} + \frac{1}{D_e} \right)$$
 (20)

But since $D_e >> D_i$ we may simplify,

$$\frac{dK}{dy} = \frac{n_{e}}{D_{i}}$$
 (21)

Figure 2 shows the behavior of K⁻ as defined in Equation 19. As can be seen from the sketch, the term has either a positive or zero slope, so we may conclude that

$$w_{e}$$
 or $n_{e} > 0$ (a) anode region
$$= 0$$
 (b) undisturbed plasma
$$> 0$$
 (c) cathode region

Of course, this means that ionization must exceed recombination by an appropriate amount in order to satisfy Equations 16-18 and 10, 11. As mentioned previously, we shall assume that ionization is due to electron impact and that recombination is of the two or three-body type as appropriate for moderately high pressure discharges.

The material presented in this section points out some interesting facts. These may be summarized by stating that the geometry of the current flow is not necessarily imposed by the electrode geometry. Depending on the level of the

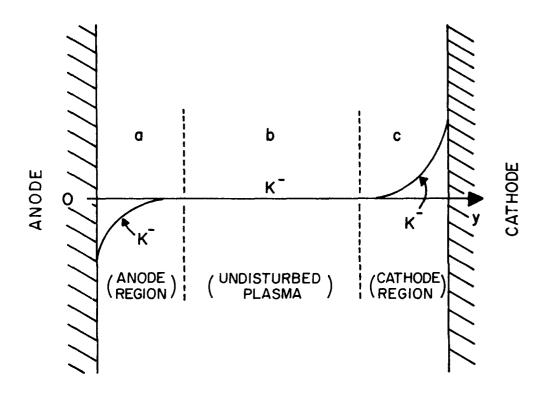


Figure 2. Behavior of K $^-$ at the Anode, Undisturbed Plasma and Cathode Regions

current, the plasma constituents, etc., we may have a spot mode or a glow mode (one-dimensional disregarding end effects) at one or both of the electrodes¹². Moreover, the sheath and ambipolar regions may grow with increasing current making the solution of the problem a challenging one. It is clear that one does not a priori specify the dimensionality and size of the domain for calculation but that one has to make certain that the description will be sufficiently unrestricted to permit a solution of the problem.

There are other instances where a given formulation is tractable only for certain geometries or shapes such as the flow of a uniform, incompressible viscous fluid due to a moving body at small Reynolds numbers (Stoke's flow)²⁷.

SECTION III

PROBLEM SOLUTION

1. Working Equations

It has been shown above that the depending on the level of the current and associated voltage, the electrode regions represent a multidimensional problem. We are, therefore, modeling non-emitting electrodes as depicted in Figure 3. Here a periodic, 2-dimensional, flat plate region is seen. We assume the coefficients to be either constant as in the case of the ions and neutrals, or dependant on E/N. For the range of E/N considered, it is adequate to take the electron diffusion coefficient and the electron temperature as 18,28

$$D_e = 5.5 \times 10^{-2} \text{ (m}^2/\text{s)}$$
 (22)

$$T_e \approx \frac{12.1\ln(E/Nx10^{+20}) + 38.5}{38.5} T_{e_{\infty}}^{(0)} K)$$
 (23)

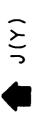
Note that E/N is in $Volts-m^2$ in the above.

Since we are neglecting convection and assuming that the sustainer operates in the afterglow of the E-beam current, the governing equations in our model are written as follows

$$-\hat{\nabla} \cdot \left[\beta \hat{\mathbf{n}}_{\mathbf{i}} \hat{\nabla} \hat{\phi} + \hat{\nabla} \hat{\mathbf{n}}_{\mathbf{i}}\right] = \frac{\mathbf{D}_{\mathbf{e}}}{\mathbf{D}_{\mathbf{i}}} \hat{\mathbf{n}}_{\mathbf{e}}$$
 (24)

$$\hat{\nabla} \cdot \left[\frac{\hat{\mathbf{n}}_{e} \hat{\nabla} \hat{\mathbf{\Phi}}}{\theta} - \hat{\nabla} \hat{\mathbf{n}}_{e} - \frac{\hat{\mathbf{n}}_{e}}{\theta} \hat{\nabla} \hat{\mathbf{E}} \right]^{T} = \hat{\mathbf{n}}_{e}$$
 (25)

$$\hat{\nabla}^2 \hat{\phi} = \gamma_p (\hat{n}_e - \hat{n}_i)$$
 (26)



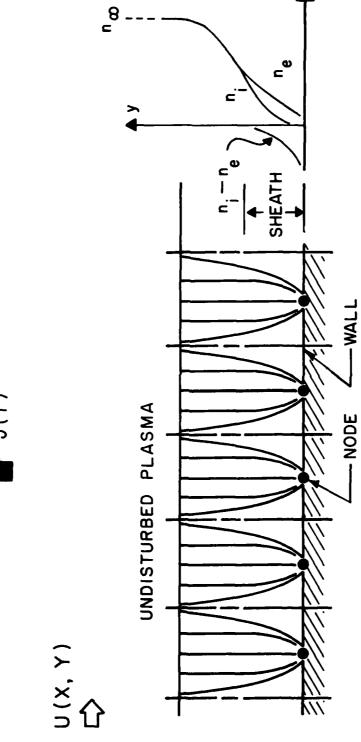


Figure 3. Anode Model with Periodic Current Constrictions

where
$$\hat{n}_{i} = C_{i}/C_{i\infty}$$
 $\hat{n}_{e} = C_{e}/C_{e\infty}$
 $\hat{\phi} = e\phi/kT_{e\infty}$
 $\theta = T_{e}/T_{e\infty}$
 $\hat{n}_{e} = \hat{w}_{e}(\lambda_{s}^{2}/C_{e\infty}\rho D_{e})$
 $\beta = T_{e_{\infty}}/T_{o}$
 $\gamma_{p} = \frac{e|\phi_{s}-\phi_{o}|}{kT_{e_{\infty}}}$

Now, we specialize our formulation to the flate plate depicted in Figure 3. We drop the carets (^) for simplicity, but it is to be understood that all variables have been suitably non-dimensionalized. In particular,

$$\gamma \beta n_{\mathbf{i}} (n_{\mathbf{i}} - n_{\mathbf{e}}) - \beta \left(\frac{\partial n_{\mathbf{i}}}{\partial \mathbf{x}} \frac{\partial \phi}{\partial \mathbf{x}} + \frac{\partial n_{\mathbf{i}}}{\partial \mathbf{y}} \frac{\partial \phi}{\partial \mathbf{y}} \right) - \left(\frac{\partial^{2} n_{\mathbf{i}}}{\partial \mathbf{x}^{2}} + \frac{\partial^{2} n_{\mathbf{i}}}{\partial \mathbf{y}^{2}} \right) = \left(\frac{D_{\mathbf{e}}}{D_{\mathbf{i}}} \right) n_{\mathbf{e}}$$

$$\gamma n_{\mathbf{e}} (n_{\mathbf{e}} - n_{\mathbf{i}}) \theta^{-1} + \left(\frac{\partial n_{\mathbf{e}}}{\partial \mathbf{x}} \frac{\partial \phi}{\partial \mathbf{x}} + \frac{\partial n_{\mathbf{e}}}{\partial \mathbf{y}} \frac{\partial \phi}{\partial \mathbf{y}} \right) \theta^{-1} - \theta^{-2} n_{\mathbf{e}} \left(\frac{\partial \phi}{\partial \mathbf{x}} \frac{\partial \theta}{\partial \mathbf{x}} + \frac{\partial \theta}{\partial \mathbf{y}} \frac{\partial n_{\mathbf{e}}}{\partial \mathbf{y}} \right) - \left(\frac{\partial^{2} n_{\mathbf{e}}}{\partial \mathbf{x}^{2}} + \frac{\partial^{2} n_{\mathbf{e}}}{\partial \mathbf{y}^{2}} \right)$$

$$+ \theta^{-2} n_{\mathbf{e}} \left[\left(\frac{\partial \theta}{\partial \mathbf{x}} \right)^{2} + \left(\frac{\partial \theta}{\partial \mathbf{y}} \right)^{2} \right] - \theta^{-1} n_{\mathbf{e}} \left(\frac{\partial^{2} \theta}{\partial \mathbf{x}^{2}} + \frac{\partial^{2} \theta}{\partial \mathbf{y}^{2}} \right) = \dot{n}_{\mathbf{e}}$$

$$(28)$$

$$\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = \gamma (n_e - n_i)$$
 (29)

The associated boundary conditions are shown in Figure

4. Each node is in a periodic field whose side or x-dimension is given by a few sheath lengths. This "packing" is arbitrary but yields a maximum total current at the electrode for a given electrode voltage drop. The latter is governed by the individual node current. Thus, we model a maximum "crowding" condition.

Any further crowding would precipitate the same situation that negates the one-dimensional solution. The top or y-dimension is unbounded, as required by the boundary-layer type of behavior. Note, furthermore, from Figure 4 that the electron density at the anode is not zero as is often done in conventional probe analysis. Such a change is needed because a finite node current requires a finite node charge density. The node density is found by trying various values until a match of the current at the node and the current at the undisturbed plasma is obtained.

The continuous domain around one node is modeled with equidistant grid points on a square mesh as shown in Figure 4. The finite difference equations approximating the two species equations and Poisson's equation are solved simultaneously in a line-iterative fashion.

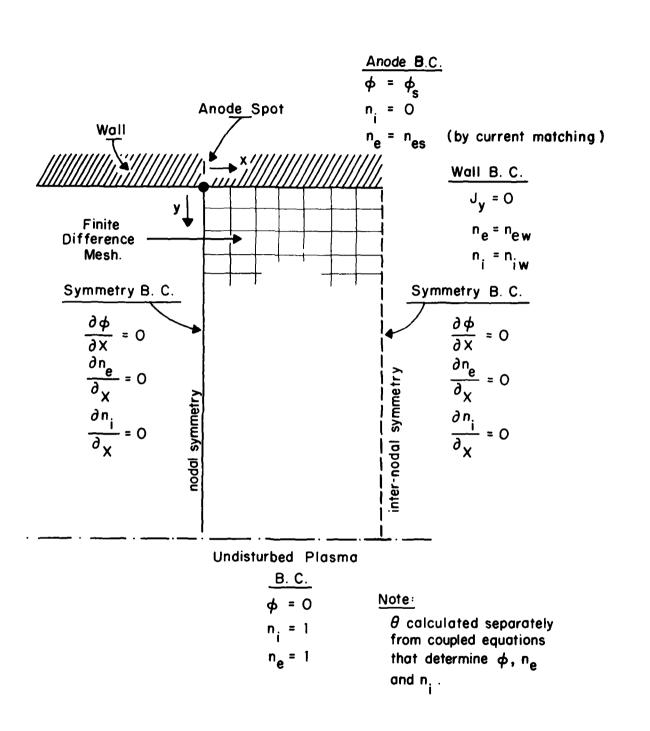


Figure 4. Boundary Conditions for Computational Method

2. Program

The electron and ion species equations (Equation 27 and 28) contain non-linear terms which present problems in their computer analsyis. The Jacobi method includes all non-linear terms on the "right hand side", i.e., external to the coefficient matrix. Convergence to a solution is possible if these non-linear terms change slowly enough over each iteration. was found by Dolson in Reference 29 that the Jacobi method was in fact unstable for the present set of equations and conditions. As a consequence of the failure of the Jacobi method, a quasi-Jacobi method was used, in which an estimate was computed for each total variable (ϕ , n_e , n_i) at each grid point during each iteration. When the product of two variables was encountered, one variable was treated as a constant coefficient for each iteration. This means that the non-linear terms are retained in the coefficient matrix. The "constant" coefficients are updated after every iteration, thus changing the coefficient matrix. The conventional Jacobi method was found to converge only for very low values of \$\phi\$ at the electrode, whereas the quasi-Jacobi procedure provided converged solutions for values of ϕ < 2 volts. Thus, the Newton-Raphson method was chosen.

The Newton-Raphson method, Reference 30, is presented by way of illustration. Equations 27, 28 and 29 are the three non-dimensional equations for ϕ , n_e , and n_i . The Poisson Equation can be written at grid node i,j as:

$$\frac{\partial^2 \phi}{\partial \mathbf{x}^2} + \frac{\partial^2 \phi}{\partial \mathbf{v}^2} + \gamma (\mathbf{n_i} - \mathbf{n_e}) = \mathbf{F_{1,ij}}$$
 (30)

The solution in this case is said to have converged ideally if $F_{1,ij} = 0$ or practically if the k^{th} iteration value $F_{1,ij}^k < \epsilon$. Writing this equation in a linear truncated Taylor series expansion, and dropping i,j subscripts for simplicity:

$$F_{1}^{k+1} = F_{1}^{k} + \Delta F_{1}^{k+1}$$
where
$$\Delta F_{1} = F_{1}(\Delta \phi, \Delta n_{e}, \Delta n_{i})$$
and
$$\phi^{k+1} = \phi^{k} + \Delta \phi^{k+1} \quad etc.$$
or
$$\frac{\partial^{2} \phi^{k+1}}{\partial x^{2}} + \frac{\partial^{2} \phi^{k+1}}{\partial y^{2}} + \gamma (n_{i}^{k+1} - n_{e}^{k+1}) =$$
(32)

$$\frac{\partial^2 \phi^{\mathbf{k}}}{\partial \mathbf{x}^2} + \frac{\partial^2 \phi^{\mathbf{k}}}{\partial \mathbf{y}^2} + \gamma (\mathbf{n_i}^{\mathbf{k}} - \mathbf{n_e}^{\mathbf{k}}) + \frac{\partial^2 \Delta \phi}{\partial \mathbf{x}^2} + \frac{\partial^2 \Delta \phi^{\mathbf{k}+1}}{\partial \mathbf{y}^2} + \gamma (\Delta \mathbf{n_i}^{\mathbf{k}+1} - \Delta \mathbf{n_e}^{\mathbf{k}+1})$$

This equation must hold at each of the ixj grid points of the finite difference mesh.

A typical non-linear term:

$$\nabla \phi \cdot \nabla \mathbf{n}_{\mathbf{e}} = \frac{\partial \phi}{\partial \mathbf{x}} \frac{\partial \mathbf{n}_{\mathbf{e}}}{\partial \mathbf{x}} + \frac{\partial \phi}{\partial \mathbf{y}} \frac{\partial \mathbf{n}_{\mathbf{e}}}{\partial \mathbf{y}}$$
(33)

The expanded iteration form would look like, (x-terms only):

$$\frac{\partial \phi}{\partial \mathbf{x}} = \frac{\partial \mathbf{n}_{\mathbf{e}}}{\partial \mathbf{x}} = \frac{\partial \phi}{\partial \mathbf{x}} = \frac{\partial \phi}{\partial \mathbf{x}} = \frac{\partial \mathbf{n}_{\mathbf{e}}}{\partial \mathbf{x}} + \frac{\partial \mathbf{n}_{\mathbf{e}}}{\partial \mathbf{x}} + \frac{\partial \mathbf{n}_{\mathbf{e}}}{\partial \mathbf{x}} + \frac{\partial \phi}{\partial \mathbf{x}} = \frac{\partial \phi}{\partial \mathbf{x}} = \frac{\partial \phi}{\partial \mathbf{x}} = \frac{\partial \phi}{\partial \mathbf{x}} + \frac{\partial \phi}{\partial \mathbf{x}} = \frac{\partial \phi}{\partial \mathbf{x}}$$

 ΔF is considered linear in $\Delta \varphi$, Δn_e and Δn_i and so the last term in equation 34, is neglected, and the products of the k^{th} solutions are known, as are the k^{th} coefficients of the unknowns, $\Delta \varphi^{k+1}$ and Δn_e^{-k+1} .

Thus the final matrix has the form:

kth Coefficients

$$\begin{bmatrix} k \\ n_{e}^{k}, \\ n_{e}^{k}, \\ n_{i}^{k} \end{bmatrix} \begin{bmatrix} \Delta \phi^{k+1} \\ \Delta n_{e}^{k+1} \\ \Delta n_{i}^{k+1} \end{bmatrix} = -\begin{bmatrix} F_{1} \\ F_{2} \\ F_{3} \end{bmatrix}^{k}$$
(35)

where the F_i 's correspond in form to Equation 30, for each of Equations 27, 28 and 29.

Thus the solution procedure consists of evaluating the right hand side based upon the \mathbf{k}^{th} solution and solving the system of equations for $\Delta \boldsymbol{\varphi}^{k+1}$, Δn_e^{-k+1} and Δn_i^{-k+1} which are used to update their respective grid point values. Appendix C shows the program listing with additional detail.

SECTION IV

RESULTS

Computer solutions were obtained for the set of conditions presented in Table I. The anode voltage is 10 volts in all cases.

TABLE 1: COMPUTER SOLUTIONS CONDITIONS

CASE	'nе	9	Н
I	not coupled	not coupled	2.0
ΙΙ	coupled	not coupled	2.0
III	coupled	not coupled	0.5
ΙV	coupled	coupled	0.5

The results are shown in Figures 5-8 as three-dimensional views of the potential field, temperature field, space charge density, electron and ion densities. Also included are graphs of potential, species and space charge densities along a cut from the anode to the free stream for each case.

Table I gives the five case conditions, where $\dot{n}_{\rm e}$ represents the "right hand side" of Equations (28) and (29), θ represents the temperature equation which is a function of the solution set (E/N, properly) and H is equal to the size of the computational node spacing, i.e., $H\lambda_{\rm S}$ is physical space.

Initial solutions were obtained with $\dot{n}_e=0$ with the sheath size smaller than the grid spacing. These coarse solutions were illustrative of the ambipolar solution in which $\nabla^2\phi \approx 0$ except at the anode where the space charge is forced as a "boundary condition". The effect of "turning-on" the

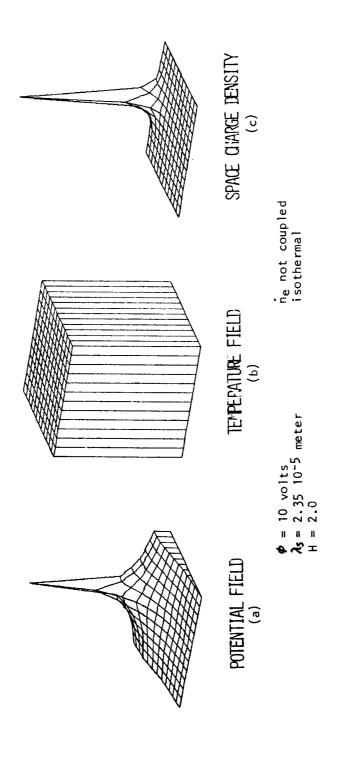


Figure 5. Three dimensional presentations of CASE I. (a) potential field, (b) temperature field, and (c) space charge density.

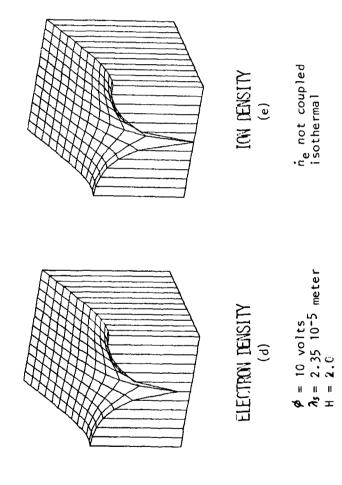
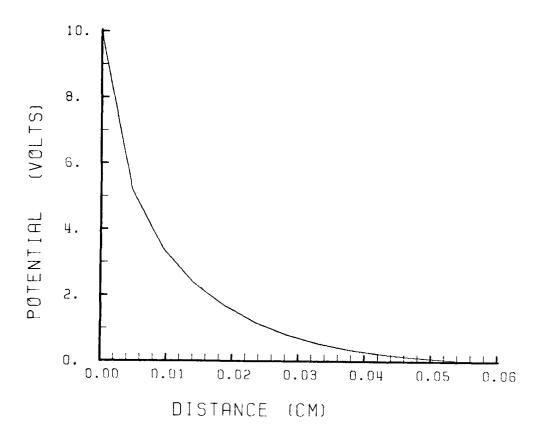
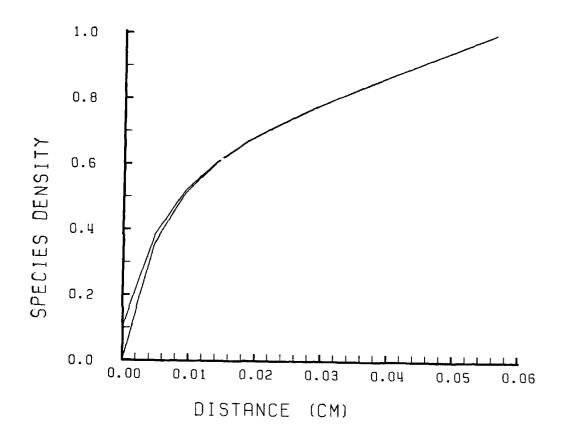


Figure 5 (continued). Three dimensional presentations of CASE I. (d) back view of electron density and (e) backview of ion density.



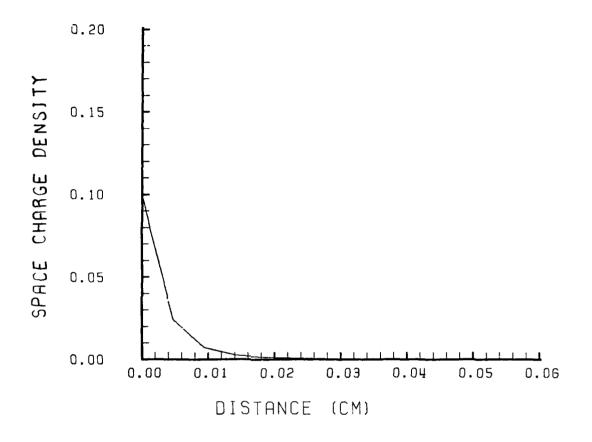
 ϕ = 10 volts λ_5 = 2.35 10⁻⁵ meter H = 2.0 n_e not coupled isothermal

Figure 5 (f). Two dimensional presentation of potential vs. distance from the anode to the freestream.



 ϕ = 10 volts λ_5 = 2.35 10⁻⁵ meter H = 2.0 he not coupled isothermal

Figure 5 (g). Two dimensional presentation of electron and ion densities vs. distance from the anode to the freestream.



 ϕ = 10 volts λ_9 = 2.35 10⁻⁵ meter H = 2.0 ne not coupled isothermal

Figure 5 (h). Two dimensional presentation of space charge density vs. distance from the anode to the freestream.

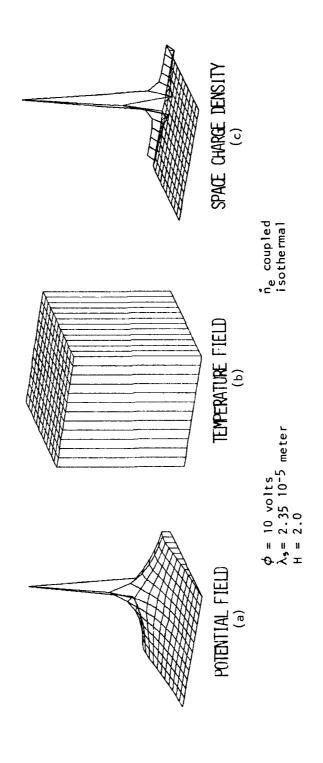


Figure 6. Three dimensional presentation of CASE II. (a) potential field, (b) temperature field, and (c) space charge density.

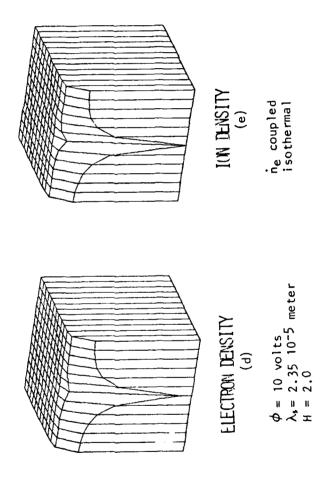
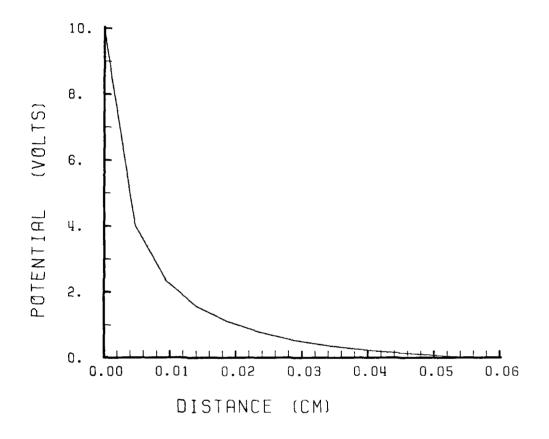


Figure 6 (continued). Three dimensional presentations of CASE 11. (d) back view of electron density, and (e) back view of ion density.



$$\phi$$
 = 10 volts \hat{n}_e coupled λ_s = 2.35 10⁻⁵ meter isothermal H = 2.0

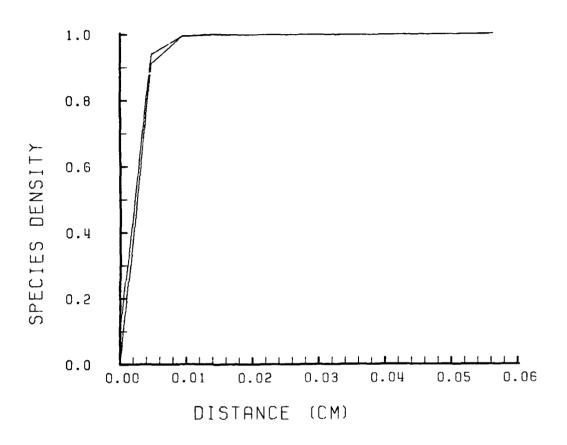
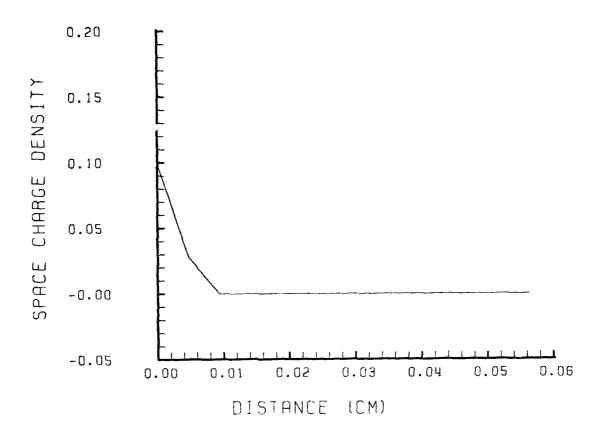


Figure 6 (g). Two dimensional presentation of electron and ion densities vs. distance from the anode to the freestream.



 ϕ = 10 volts λ_s = 2.35 10⁻⁵ meter H = 2.0 ne coupled isothermal

Figure 6 (h). Two dimensional presentation of space charge density vs. distance from the anode to the freestream.

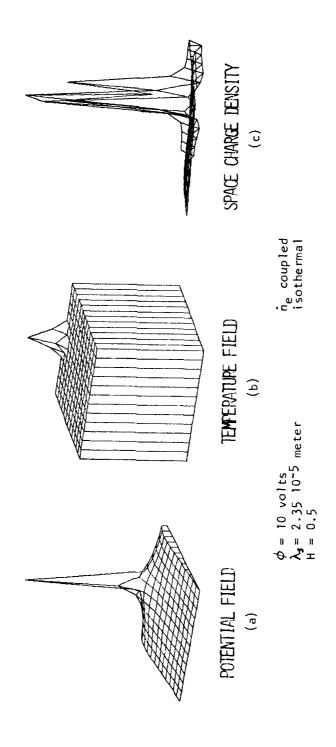


Figure 7. Three dimensional presentation of CASE III. (a) potential field, (b) temperature field, and (c) space charge density.

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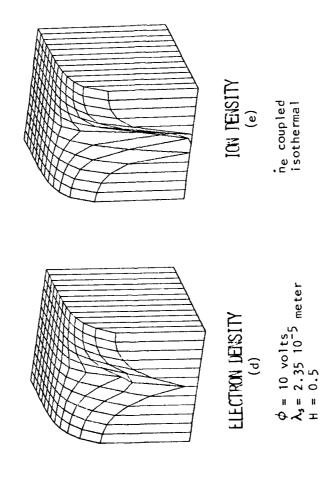
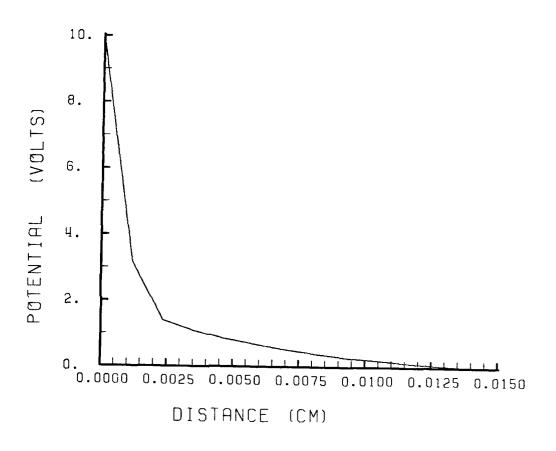


Figure 7 (continued). Three dimensional presentation of CASE III. (d) back view of electron density, and (e) back view of ion density.



$$\phi$$
 = 10 volts $\dot{\lambda}_s$ = 2.35 10⁻⁵ meter isothermal $\dot{\mu}$ = 0.5

Figure 7 (f). Two dimensional presentation of potential vs. distance from the anode to the freestream.

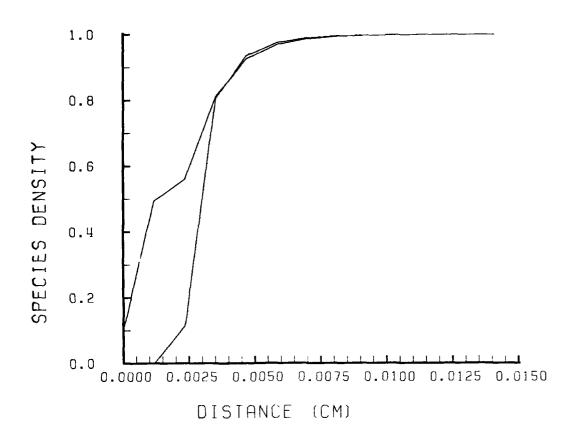
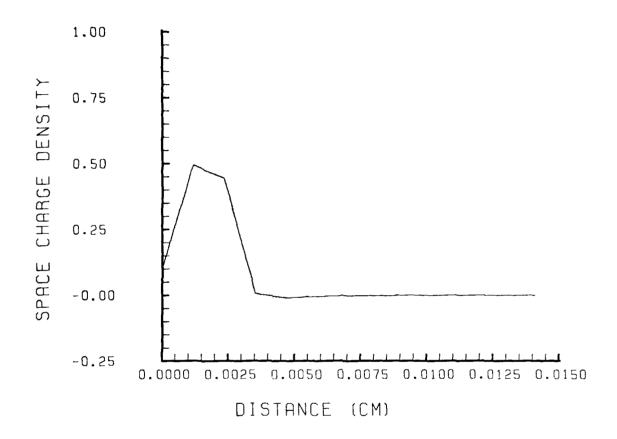


Figure 7 (g). Two dimensional presentation of electron and ion densities vs. distance from the anode to the freestream.



 ϕ = 10 volts \dot{n}_e coupled λ_s = 2.35 10⁻⁵ meter isothermal H = 0.5

Figure 7 (h). Two dimensional presentation of space charge density vs. distance from the anode to the freestream.

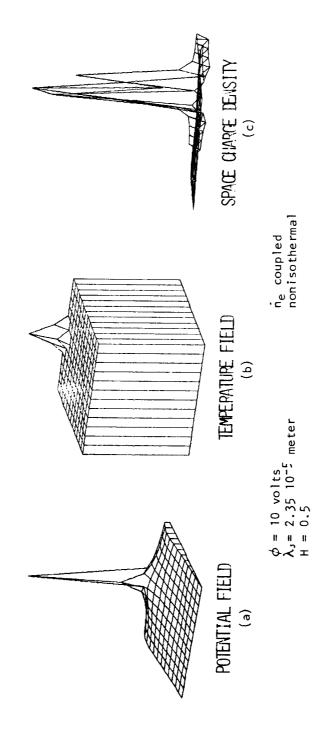


Figure 8. Three dimensional presentation of CASE IV. (a) potential field, (b) temperature field, and (c) space charge density.

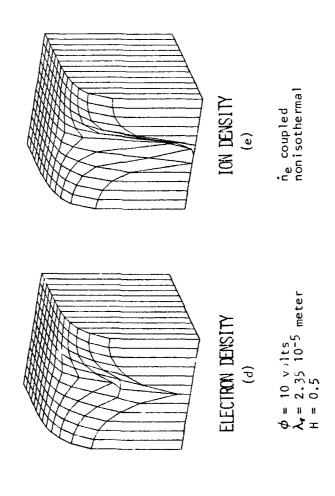
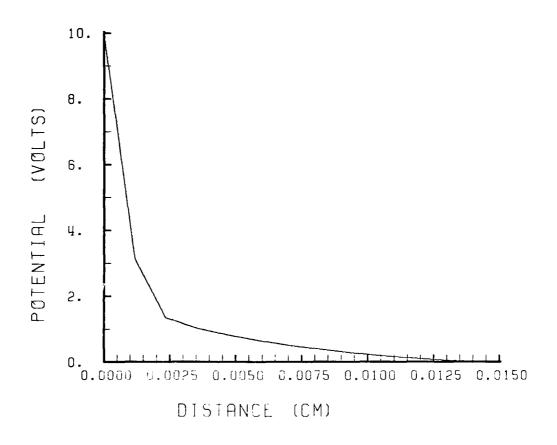


Figure 8 (continued). Three dimensional presentation of CASE IV. (d) back view of electron density, and (e) back view of ion density.



$$\phi$$
 = 10 volts \dot{n}_e coupled \dot{n}_s = 2.35 10⁻⁵ meter nonisothermal H = 0.5

Figure 8 (f). Two dimensional presentation of potential vs. distance from the anode to the freestream.

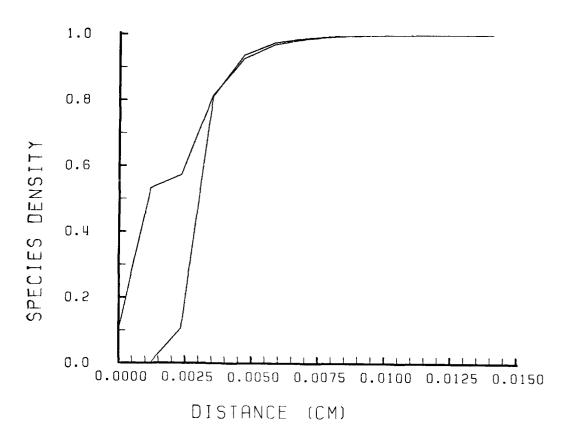
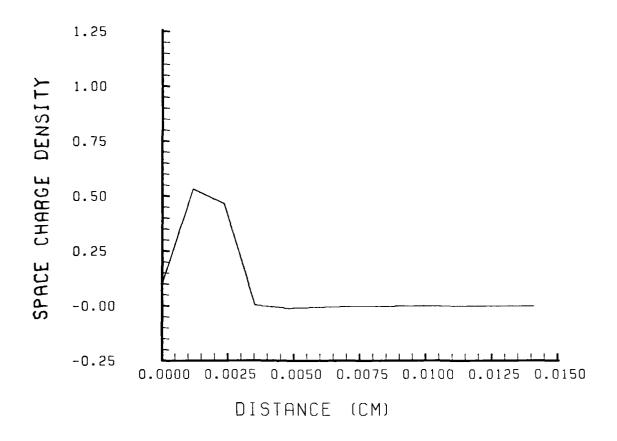


Figure 8 (g). Two dimensional presentation of electron and ion densities vs. distance from the anode to the freestream.



$$\phi$$
 = 10 volts $\dot{n}_{\rm e}$ coupled $\lambda_{\rm s}$ = 2.35 10⁻⁵ meter nonisothermal H = 0.5

Figure 8 (h). Two dimensional presentation of space charge density vs. distance from the anode to the freestream.

production term (h_e) is to bring about the boundary layer behavior of the solution (Case II). The boundary layer behavior of the species densities is easily illustrated in the comparison of Figure 5(g) (Case I) to Figure 6(g) (Case II). Case I and II conditions produced an insufficient electric field to enable the electron temperature θ to be greater than 1. Thus Case I and II conditions were identical with the temperature routine coupled or uncoupled.

A solution would always be obtained as long as the sheath is not larger than the grid size spacing, for $\dot{n}_e=0$. However reasonable solutions could be obtained for grid size spacing smaller than the sheath size only for $\dot{n}_e\neq0$. Cases III and IV each represent the sheath as 2 grid spacing ($\mu=0.5$). Case IV is identical to Case III with the addition of the electron temperature effects. The temperature effect in Case IV produced only a slight increase of space charge density and a very small increase of the electron field near the anode.

In all cases the current into the inactive wall was 3 or 4 orders of magnitude less than the current at the anode node. The current at the anode node was approximately 3 times the total current at the free stream. The anode current could be better matched to the free stream current by picking a suitable $n_{\rm e}$ (anode), but this was not pursued because previous results of a preliminary nature indicate only minor changes over the results shown. In fact in cases I and II with $n_{\rm e}$ = 0 , at the anode, the anode current still exceeds the current at the

free stream boundary. The electric field at the free stream boundary is insufficient to offer a current matching condition. A more proper free stream boundary condition would be $E=E_{\infty}$ rather than $\varphi=0$. The electron field boundary condition created numerical instabilities and is recommended as a subject for later analysis. Our approach thus far was to increase φ_O large enough to produce the proper E_{∞} .

SECTION V

SUMMARY AND CONCLUSIONS

The results presented in this report are based on a two-dimensional description of the current flow at the anode. Diffusion is properly accounted for as is ionization/recombination. No convection or magnetic field effects are included. We assume operation in the after-glow of the E-beam ionization so that ψ and q are essentially uncoupled from the solutions. Current constricts at active nodes along the surface, and the sheath and ambipolar region are self-generating in the solution of Equations 27-29.

The boundary layer nature of the sheath is clearly evident in the results shown; the boundary layer nature of the ambipolar region only shows up when \dot{w}_e or \dot{n}_e are coupled into the calculations. The magnitude of \dot{n}_e is highest within the sheath and along the walls, dropping off towards the undisturbed plasma. The space charge density (n_e-n_i) peaks at $0.5 \times 10^{18} \text{m}^{-3}$ near the active node.

About 90% of the voltage drop takes place within one or two sheath lengths, this produces a maximum electric field of 0.8×10^6 V/m (for the 10 volts drop) at the electrode node. The undisturbed region electric field is about 0.5×10^4 V/m which is considerably below the value of 0.27×10^6 V/m required for pumping the laser medium. But then again, ϕ is 10 V and it should be 35 V. That is, roughly, an increase by a factor is 54 required for E. The temperature field, being a direct

function of E/N, does not affect our results simply because E/N is yet too low. There are, however, subtle changes which we feel are going to become important at the higher voltages.

It is anticipated that H can be decreased further (up to 1/6 of a sheath length), with the coupling of \dot{n}_e , and that this will allow smoother changes within the sheath and more numerical stability at higher voltages. Also, it should be possible to consider jointly the E-beam ionization in our solutions to produce more realistic effects.

APPENDIX A

AMBIPOLAR REGION -- ONE DIMENSIONAL DESCRIPTION

We have seen that within the sheath, the solution to the conservation equations cannot be one dimensional. While there may be some particular range of ionization/recombination which permits one-dimensional solutions, most often we must allow for current concentrations at the electrodes. However, away from the electrodes - the problem may become one dimensional in the ambipolar region (see Figures 3, Section III.1). Since the extent of the sheath is quite small when compared to typical interelectrode dimensions, it is interesting to examine one-dimensional solutions within and beyond the ambipolar region. Here we assume that an initial concentration, n_s, is known at the wall-end of the ambipolar region and that a symmetry plane exists for the charge concentration (usually the mid-plate region), see Figure Al.

Equations A1, A2, and A5 are shown below for the one-dimensional, ambipolar region. We use number density instead of mass fraction and take S to represent the E-beam production rate of secondary electrons. Convection is neglected and the resulting electric field in this region is constant. Also, all the coefficients turn out to be constant.

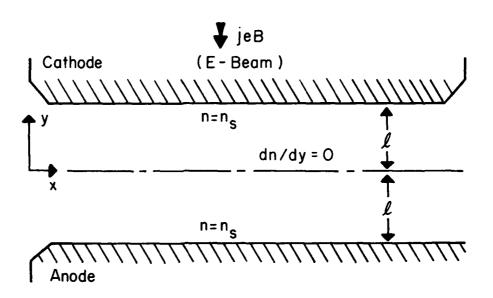


Figure Al. E-Beam Discharge Configuration

$$\frac{d}{dy} \left[\frac{eP_i}{kT_O} rE - D_i \frac{dn}{dy} \right] = \dot{n}_e + S$$
 (A1)

$$\frac{d}{dy} \left[\frac{-eD_e}{kT_e} nE - De \frac{dn}{dy} \right] = \dot{n}_e + S + o$$
 (A2)

$$dE/dy = 0$$
 or $E = constant$ (A3)

We now study Equation Al in the afterglow region. Let S and q be zero so that

$$\dot{n}_e = v_i^n - \alpha n^2 \tag{A5}$$

and combine with Equation Al.

$$\frac{D_{i}}{v_{i}} \left[\frac{eE}{kT_{o}} \frac{dn}{dy} - \frac{d^{2}n}{dy^{2}} \right] = n - n^{2}/n*$$

In order to nondimentionalize the above relation we introduce the ambipolar diffusion characteristic length along with suitable ratios of the variables.

$$L_{n} = \sqrt{D_{i}/v_{i}} \text{ or } L_{n}N = \sqrt{\frac{D_{i}N}{v_{i}/N}}$$

$$\hat{y} = y/L_{n}$$

$$\hat{n} = n/n*$$

$$\hat{E} = \frac{e^{F}L_{n}}{kT_{0}} = \frac{e(E/N)L_{n}N}{kT_{0}}$$
(A6)

The resulting differential equation becomes

$$\frac{d^2\hat{n}}{d\hat{y}^2} - \hat{E} \frac{d\hat{n}}{d\hat{y}} + (\hat{n} - \hat{n}^2) = 0$$

$$(d\hat{n}/d\hat{y})_{y=0} = 0 \quad \text{and} \quad \hat{n}(0) = \eta$$

$$\hat{n}(\pm \ell) = \hat{n}_{g}$$
(A7)

It is obvious from equation A7 that if $\hat{n}'(0)=0$ and $\hat{n}(0)=1$ we have truly boundary-layer behavior since both \hat{n}' and \hat{n}'' become zero at the midplane. This represents a mathematical limit but note that because L_n is very small in relation to the interelectrode distance, $\hat{n}(0)$ will be for all practical purposes equal to one, i.e., $n\approx 1.0$. Figure 1 in the text shows calculated values of L_nN , as a function of E/N. These compare reasonably well with values quoted in the literature.

Equation A7 is written below without the carets and with primes instead of derivatives.

$$n'' - En' + (n-n^2) = 0$$
 (A8)
 $n'(0) = 0$
 $n(\pm \ell) = n_s$

We have considered the above equation for E \geqslant 51 (or E/N \geqslant 10^{-16}V-cm^2). The solution of Equation A8 was obtained

numerically and is shown in Figure A2. We note that this solution can be, within a very close approximation, easily obtained by neglecting n''. We have

$$n' \approx \frac{1}{E}(n-n^2) \tag{A9}$$

and $n' \approx 0$ when n = 1

The important result is that $n' \approx 0(1/E)$ and $n'' \approx 0(1/E^2)$ so that only for E sufficiently large may we neglect diffusion in the ambipolar region.

Solving Equation A9 we obtain:

$$n = \left[1 + (1/n_s^{-1}) \exp\left(\frac{y-\hat{x}}{E}\right)\right]^{-1}$$
 (A10)

Now, since $\hat{\ell} >> \hat{E}$ due to the smallness of L_n relative to $\hat{\ell}$ (i.e., $L_n > 10^{-2} \text{cm}$ whereas $\ell > 10 \text{cm}$), $\hat{n}(0) \approx 1$; furthermore $n(\ell) = n_s$. For values $\hat{n}_s > 0.5$, it is easy to see that the above approximation is valid. Results shown in a previous section indicate that this is reasonably true. Thus, diffusion is unimportant here.

In dimensional form, Equation AlO becomes

$$\frac{n}{n^*} = \left[1 - \left(\frac{n^*}{n_S} - 1\right) \exp\left(\frac{y - \ell}{eEL_n^2/kT_O}\right)\right]^{-1}$$

$$n^* = v_i/\alpha$$

$$L_n = \sqrt{D_i/v_i}$$
(A11)

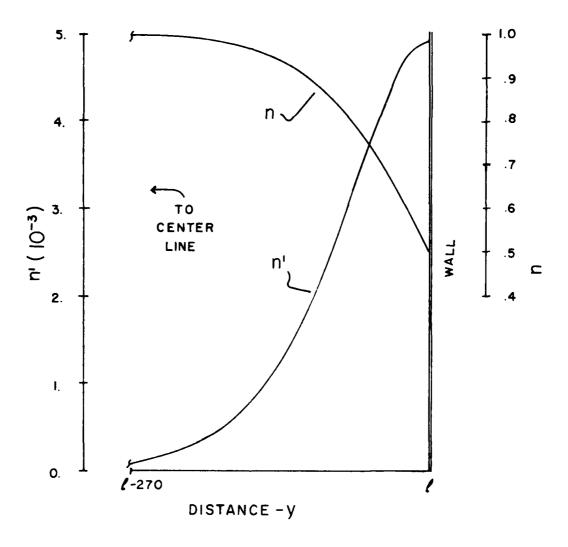


Figure A2. n and n' versus Spacing

If we now proceed to re-examine Equations A1 and A2, assuming that the magnitude of n" and q are insignificant but that the E-beam is operating, we have

$$\frac{eD_{i}E}{kT_{i}}\frac{dn}{dy} \approx \dot{n}_{e} + S$$
 (A12)

$$-\frac{e^{D}e^{E}}{kT_{e}}\frac{dn}{dy} \approx \dot{n}_{e} + S$$
 (A13)

Clearly, the above set of equations has only a trivial, nonphysical solution. We see again that a one dimensional forulation is not feasible.

It was mentioned in Section II-1 that $\dot{n}_e \geqslant 0$ was a requirement for stability in the electrode regions. We now briefly discuss what this implies in the presence of the E-beam. Recall

$$\dot{n}_e = v_i - \alpha n^2 + S \tag{A14}$$

Let

$$n_0^2 = S/\alpha$$

So that
$$\dot{n}_e = \alpha (n*n + n_0^2 - n^2) \ge 0$$
 (A15)

Solving for n, $\alpha \neq 0$,

$$n \le \frac{1}{2}n^* + \sqrt{\frac{n^{*2}}{4} + n_0^2}$$
 (A16)

Now n* is a function of E/N as seen from Figure 1 and typically n* is much less than n_{Ω} away from the electrode

regions. According to Equation A16, potential difficulties can be encountered in regions where S is sufficiently small so that as E/N and n* decrease away from the electrode (i.e., as the discharge becomes non-self-sustaining), $\dot{n}_{\rm e} < 0$. This is because n can exceed n locally at the fringes of the E-beam Whether or not such a thing can trigger arcing remains to be established.

APPENDIX B

MORE ON PROBLEM DIMENSIONALITY

The arguments introduced in Section II.3 relating to the dimensionality Equations 11-13 are based on physical grounds. Here, we present two more mathematical proofs which, while not sufficient, show clearly the inappropriateness of the resulting solutions. It is possible to combine Equations 16-18 into the single equation for the electric field seen below

$$\frac{kT_{O}\varepsilon_{O}}{e^{2}E}(\frac{d^{3}E}{dy^{3}}) - \frac{kT_{O}\varepsilon_{O}}{2e^{2}E^{2}}\frac{d}{dy}\left[\frac{dE}{dy}^{2}\right] - \frac{kT_{O}}{eE^{2}}K^{+}\left(\frac{dE}{dy}\right) - \frac{\varepsilon_{O}}{2kT_{O}}\frac{d}{dy}(E^{2}) + K^{-} = 0$$
(B1)

where
$$K^+ = j_i/eD_i + j_e/eD_e$$
 and $K^- = \frac{j_i}{eD_i} - \frac{j_e}{eD_e}$

From Reference 12
$$\delta = n_i - n_e = \frac{\epsilon_0}{e} dE/dy$$

and

$$\sigma = n_i + n_e = \frac{kT_o}{eE} (\frac{d\delta}{dy} + K^+)$$

Equation B1 is a third order, ordinary non-linear differential equation. We may nondimensionalize it with

$$E_{O} = \frac{kT_{O}}{eL} \qquad E \leftrightarrow E/E_{O}$$

$$L = \frac{e^{2}}{\epsilon_{O}kT_{O}} \qquad y \leftrightarrow y/L$$
(B2)

From a dimensional analysis of Equation Bl , it is found that the above substitutions will result in a form of the equation free of coefficients. Equation Bl thus becomes,

substituting primes for the derivatives

$$E''' - \frac{1}{2E}(E'^2)' - \frac{K}{E}E' - \frac{E}{2}(E^2)' + EK^- = 0$$
 (B3)

Where, in the above, E, K, and y are nondimensional. We can look for solutions to this equation which monotonically decrease from the electrode to a small value at the undisturbed plasma.

A solution which is both very simple and reasonable for the anode is

$$E(y) = 2(y + a)^{-1}$$
 (B4)
 $a = 2/E_a$

This solution satisfies Equation B3 for the special case

$$K^{+} = -2K^{-} \tag{B5}$$

or

$$j_{i} = \frac{1}{3} \frac{D_{i}}{D_{e}} j_{e}$$
 (B6)

Moreover,

$$E' = \delta = n_i - n_e$$

= $-2(v + a)^{-2}$ (B7)

is also reasonable for the anode since the space charge is negative and rapidly vanishes as y grows. Now the sum of the charge densities is found from

$$\sigma = n_i + n_e = \frac{1}{E} \left[K^+ + \frac{d\delta}{dy} \right]$$
 (B8)

$$= \frac{y+a}{2} \left[K^{+} + 4(y+a)^{-2} \right]$$
 (B9)

But this result is unreasonable since σ must be zero (or near zero) at the electrode and increase monotonically to an asymptotic value at the undisturbed plasma. The result shown in Equation B9 is clearly not a physically acceptable one.

The solution for the cathode is similarly obtained with

$$E = 2(a-y)^{1}, y < a$$
 (B10)

and
$$K^{+} = 2K^{-}$$
 (B11)

Please note that the cathode is to the right of the anode, as indicated in Figure ? of the text. In the above solution, both E and δ appear to be reasonable (the space charge being positive at the cathode). But, again, σ does not follow the physically acceptable pattern.

A more complete solution to Equation B3 may be obtained if we note that since K^- is zero in the undisturbed plasma, it follows that it must be zero everywhere as long as we insist on a one-dimensional, constant property solution to Equations 16-18. We have, multiplying P3 by E,

$$EE''' - \frac{1}{2}(E'^2)' - K^+E' - \frac{E^2}{2}(E^2)' = 0$$
, $K^+ = 2j_e/eD_e$ (B12)

with $E \to E_{\infty}$, $E' \to 0$, $E'' \to 0$ as $y \to the undisturbed plasma.$

Now EE''' = $(EE'')' - \frac{1}{2}(E'^2)'$

and $\frac{E^2}{2}(E^2)' = \frac{1}{4}(E^4)'$

substituting into B12

$$[EE'' - (E'^2 - K^+E - \frac{1}{4}E^4)] = 0$$

Integrating

E'' -
$$(E')^2 - K^+E - \frac{1}{4}E^4 = C_1$$
 (B13)

We evaluate C_1 at $y \rightarrow$ the undisturbed plasma

$$C_1 = -K^+ E_{\infty} - \frac{1}{4} E_{\infty}^4$$

thus

$$EE'' - (E')^2 - K^+(E-E_{\infty}) - \frac{1}{4}(E^4-E_{\infty}^4) = 0$$
 (B14)

After some algebraic manipulation we find expressions for E' and E'',

$$E' = \left[K^{+}(-2E + E_{\infty} + \frac{E^{2}}{E_{\infty}}) + \frac{1}{4}(E^{4} + E_{\infty}^{4} - 2E_{\infty}^{2}E^{2}) \right]^{\frac{1}{2}}$$
 (B15)

$$E'' = \frac{1}{2} \left[2K^{+}(-1 + \frac{E}{E_{m}}) + (E^{3} - E_{\infty}^{2}E) \right]$$
 (B16)

 $E''' = \frac{1}{2} \left[2K^{+}/E_{\infty} + (3E^{2} - E_{\infty}^{2}) \right] E', \text{ as } E \rightarrow E_{\infty}, E' \rightarrow 0 \text{ and } E''' \rightarrow 0 \text{ as well.}$

It is possible to solve for σ at this point

$$\sigma = \frac{1}{E} [E'' + K^{+}]$$

$$= \frac{K^{+}}{E_{\infty}} + \frac{1}{2} (E^{2} - E_{\infty}^{2})$$
(B17)

Furthermore, $K^+ = \sigma_{\infty} E_{\infty}$ so that

$$\sigma = \sigma_{\infty} + \frac{1}{2} (E^2 - E_{\infty}^2)$$
 (B18)

Clearly, since $E \geq E_{\infty}$, $\sigma \geq \sigma_{\infty}$ which is an improper behavior for the sum of the charge densities. Again, while δ and E seem to behave reasonably, the one-dimensional solution is incorrect for σ .

APPENDIY C

COMPUTER PROGRAM

This appendix includes the program listing with some preliminary comments.

As previously discussed the matrix form of the iteration scheme is: \mathbf{k}^{th} (oefficients

Matrix
$$\begin{bmatrix} k \\ n_{e} \\ n_{i} \end{bmatrix} \begin{bmatrix} \Delta_{i} \\ k+1 \\ \Delta n_{e} \\ k+1 \end{bmatrix} = \begin{bmatrix} F_{1} \\ F_{2} \\ F_{3} \end{bmatrix}^{k}$$
(C1)

There are three equations to be solved at each (i,j) grid point corresponding to F_1 , F_2 and F_3 (equations 27, 28 and 29). The solution vector is interlaced as:

This method was chosen to keep the values of neighboring points in the two dimensional grid mesh near each other in the solution vector. Since a grid point value F_{ij} can be expressed as a linear combination of nearest neighbor $\phi(i\pm 1,j\pm 1)$, $n_e(i\pm 1,j\pm 1)$, or $n_i(i\pm 1),j\pm 1)$ this would produce a banded coefficient matrix A. of width 46 and length 273 (for a 7x13 grid mesh) when economically stored. The International Mathematics and Statistics Library (IMSL) subroutine. LEQTIB, is suited for solving equations of this form.

The Sheath program calculates the coefficients of the matrix A and vector F (refered to as vector C in the computer program). Thus $\mathbf{C}(3n-2)$ is the \mathbf{F}_1^k value at grid mesh (i,j),

C(3n-1) is the value of the F_2^k value at grid mesh i,j etc., (where n=i+j-1). The solution to the matrix equation C_1 [$\Delta \phi$ (i,j), etc.] is assigned to vector C upon return from LEQTIB subroutine.

The solution set ϕ_i , n_e , and n_i are updated per Equation 31. The electron energy term θ_{ij} is programmed as an empirical function of E/N at grid point (i,j) and is calculated at the completion of each iteration. The iteration sequence is repeated as many times as required to converge to a solution.

The program listing follows.

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EQI=COUL*PHIZ/(BOLTZ*BETA*TGAS)*HH
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A(IR, J2) = 1. --. 5*BETA*DPHDY

A(IR, J4) = 1. +-. 5*BETA*DPHDY

A(IR, J4) = 1. +-. 5*BETA*DPHDY

A(IR, J4) = 1. +-. 5*BETA*DPHDY

A(IR, J2) = 1. --. 5*BETA*DPHDX

A(IR, J4-2) = -.. 5*BETA*DPHDX

A(IR, J4-2) = -.. 5*BETA*DNIDX

A(IR, J4-2) = -.. 5*BETA*DNIDX

A(IR, J4-2) = -.. 5*BETA*EQP*NI (IC, JC)

A(IR, J5-2) = -.. 5*BETA*EQP*NI (IC, JC)

C(IR) = -(IR, J1-2)

BETA*EQP*NI (ICM, JC) +NI (ICM, JC) +NI (IC, JC)

-BETA*EQP*(NE (IC, JC) -NI (IC, JC)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     FOR
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 HERE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              ARR I VE
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        SET
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               (WASN.
Cl=DNEDX*
CZ=-NE(IC
C3=NE(IC,
C3=NE(IC,
C4=-(DNED
C4=-(DNED
C4=-(DNED
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   4507
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                                                                                                 33
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```
SET UP INTERNODAL SYMMETRY B.C.
                                                                                                                                                                                                                                                                                                                              ပ
                                                                     SET UP NODAL SYMMETRY B.C.
                                                                                                                                                                                                                                                                                                                             COMPUTE SPECTRAL NORM OF
                                                                                                                                                                                                                                                                                                            MATRIX A IS NOW COMPLETE
                                                                                                                                                                                                                    NODE CONDITIONS SET
                                                                                        I S=NSY3+M
DD 540 IR=I S, NXNM,NSY3
A(IR, J2)=0.
A(IR, J4)=2.
CONT INUE
                                                                                                                                                              1S=2*NSY3-3+M
DO 550 IR=1S,NXNM,NSY3
A(IR,J4)=0.
A(IR,J2)=2.
CONTINUE
NXNP=NXNM+M
DO 530 IR=NXNP,NXN,3
A(IR,10)=1.
IF(M.EQ.1) C(IR)=PHQ
IF(M.EQ.2) C(IR)=NEQ
IF(M.EQ.2) C(IR)=NEQ
CONTINUE
                                                                                                                                                                                                                                              C(1) = PHIB
C(M) = NED
C(M) = NID
                                                                                                                                                                                                                                                                                                                                                      NXM2=NXN-2
CPH=0.
CNE=0.
CN = 0.
D 605 K=1, NXM2, 3
K 1=K+1
K 2=K+2
CPH=CPH+C(K)*C(K)
                                                                                                                                                                                                                                                                          540
                                                                                                                                                                                                  550
                                                     530
                                                               ပပပ
                                                                                                                                     000
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```
CALCULATE THE TEMPERATURE DISTRIBUTION
SEPARATELY FROM THE MATRIX ITERATIVE SCHEME.
11 + C(K2) + C(K2)

24 + CNE + CNI

(6,802) ITER, CER, CPH, CNE, CNI

15 Y 3 + 2

EQTIB(A,NXN,NS SS,NS Y3 + 1,NXN,C,1,NXN,0,XL,1ER)
                                                                             FROM VECTOR
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   TER.GT.3) CALL PATCH(NE,0.0,1.0)
                                                                             UPDATE P, NE, NI
                                                                                                                                                                        1 PH(I + K) = C(JJ) + PH(I + K) = C(JJ) + PH(I + K) = C(JJ) + PH(I + K) CONTINUE
                                                                                                                                                                                                                                                                                                                 JJ=J+(I-1)*NSY3+1
NE(I,K)=C(JJ)+NE(I,K)
GO TO 7000
CONTINUE
                                                                                                                000 M=1,3
                                                                                                              DO 7000 M=1,3
GC TO (710,720,730
CONTINUE
DO 711 I=1, NCOLS
                                                                                                                                                                 DC 711 J=1, NSY3, 3 K=K+1
                                                                                                                                                                                                                                                                                                                                                                                                     DC 731 I=1, NCJL S
K=0
DC 731 J=1, NSY3, 3
K=K+1
                                                                                                                                                                                                                                                                                                  00 721 J=1,NSY3,3
K=K+1
                                                                                                                                                                                                                                                                         DO 721 I=1, NCOLS
K=0
 ÷05
                                                                                                                                                                                                                                720
                                                                                                                                                                                                                                                                                                                                                          730
                                                                                                                                                                                                        711
                                                                                                                                                                                                                                                                                                                                    721
                                                                                                                                                                                                                                            000
```

```
FIRST ITGRD ITERATIONS
                                                                                                                                                                                                              JM=J-I

DPX=(-3.*PH(NCOLS,J)+4.*PH(NCMI,J)-PH(NCM2,J)1/2.

IF(J.EQ.1) JM=JP

IF(J.EQ.NSYM) JP=JM

DFY=(PH(NCOLS,JP)-PH(NCOLS,JM))/2.

GRAD(NCOLS,J)=(DPX*DPX*DPY)**.5

DO 920 J=1,NSYM

DO 920 I=2,NCMI
                                                                                                                                                                                                                                                                                                    IN= 1+1

IN= 1+1

IN= 1-1

JM=J-1

JM=J-1

JM=J-1

JM=J-1

JM=J-1

JM=J-1

JP=J+1

IF(J-EQ.NSYM) JP=JM

DPX=(PH(IP,J)-PH(IN,J))/2.

CRAD(I,J)=(DPX*DPX*DPY*DPY) **.5

DQ 930 J=1,NCOLS

TH(I,J)=(12.1*ALGG(GRAD(I,J)*EQT)+BETA)/BETA

CALL PATCH(TH,I.0,99.)

CALL PATCH(TH,I.0,1.0)

90 CONTINUE

00 CONTINUE

00 CONTINUE

00 CONTINUE

00 CONTINUE
                                      COMPUTE AS ISOTHERMAL FOR
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              990
                                                                  899
                                                                                                                                                                              006
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WRITE(6,871) WRITE(6,871) WRITE(7,800)!IMIN,!IMAX,!INDT,!IGRD,!IBC,M,H,NSTAR WRITE(6,806) ITMIN, ITMAX, ITNDT, ITGRD, ITBC, M, H, NSTAR CALL JWALL(FPH,NE,NI;TH)
CALL WRITEN(NE; UPH; VOLLAGE POTENTIAL
CALL WRITEN(NE; UPH; VOLLAGE POTENTIAL
CALL WRITEN(NE; UPH; VOLLAGE POTENTIAL
CALL WRITEN(NE; UPH; III; IGN DENSITY)
CALL WRITEN(NE; III; IGN DENSITY)
CALL WRITEN(NE; III; IGN DENSITY)
CALL WRITEN(NE; III; INSPACE CHARGE DENSITY)
CALL WRITEN(NE; III; III; INDIN, ITMAX, ITNOT, ITGRD, ITBC, M,H,NST
IFNPUNCH; GTO)
WRITE(6,802)
IFNPUNCH; GTO)
WRITE(6,802)
IFNPUNCH; GTO)
WRITE(6,802)
IFNPUNCH; GTO)
CALL PUNCHN(NE; II; TEMPERATURE DISTRIBUTION)
IFNPUNCH; GTO)
CALL PUNCHN(NE; II; TEMPERATURE DISTRIBUTION)
IFNPUNCH; GTO)
CALL PUNCHN(NE; II; TEMPERATURE DENSITY
IFNPUNCH; GTO)
CALL PUNCHN(NE; II; TEMPERATURE DENSITY
IFNPUNCH; GTO)
WRITE(6,802)
IFNPUNCH; GTO)
WRITE(6,802)
IFNTMAX, GTO, ITMOT, BALANCE RESULT ENT CURR JUIPUL FINAL CALCULATE 9100 0000ပံပပပပဲ

```
DO FGRMAT(315x, F5.2, IX, F5.2, IX, A4)

20 FORMAT(315x, F5.2)

EDRMAT(315x, F5.2)

FORMAT(315x, F5.2, IX, F5.2, IX,
TCN=.5*((-NE(1,1)*(PH(2,1)-PH(1,1))+TH(1,1)*(NE(2,1)-NE(1,1)))**2

TCN=TCN**.5

TCNT**.5

TC
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           FORMAT()
FORMAT()
STOP
END
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             8888888888
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848860484
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871
871
9959
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```
H,NE,NI,TH)
*1),NE(NCOLS,1),NI(NCOLS,1),TH(NCOLS,1)
S,NSYM,ITER
DX,DPHDY, DNEDX,DNEDY,DNIDX,DNIDY,DTHDY
C,JC,ICM,ICP,JCM,JCP
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 DETERMINES THE FIRST ORDER DIFFERENCES OF THE SOLUTION SET (PHINEINITH), AT THE NODE POINT "IC,JC". TWO POINT DIFFERENCES ARE USED.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   RETURNS THE DIFFERENCES THROUGH COMMON: DIVER.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              R ED I M ENS I ON ED
                                                                                                                                                                                                                                                                                                                                 SUBROUTINE DELOP
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    SCUBRECTINE
COMMENCE

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```

SUBROUTINE RTSIDE(NE,NI,TH,NEDOT)
DIMENSION NE(NCOLS,1),NI(NCOLS,1),TH(NCOLS,1),NEDOT(NCOLS,1)
REAL*4 NE,NI,NSTAR,NEDOT
CCMMON /IPARM/ NCOLS,NSYM,ITER
COMMON /IPARM/ NCOLS,NSYM,ITER
COMMON /IDNIZ/ RECOM,A1,A2,A3,NSTAR,T
DO 10 J=1,NCOLS
NEDOT(I,J)=NCOLS
NEDOT(I,J)=NE(I,J)*(NSTAR-NI(I,J))*RECOM
+A2*(TH(I,J)=NE(I,J)*(NSTAR-NSTAR-NE(I,J)*NI(I,J))*NE(I,J)
RETURN
RETURN CALCULATE THE NEDOT TERM OVER THE FIELD 2 9 0D Y 3 8 0D Y (HEAVY) 3 80D Y (HEAVY) 3 80D Y (ELECT)... HIN & HIRSCH SUBROUTINE RTSIDE USED TO 10

```
SUBROUTINE JWALL (PH,NE,NI,TH)

REAL*4 LAMS; JX(7)

DIMENSION PH(NCOLS,1),NE(NCOLS,1),TH(NCOLS,1)

REAL*4 NE,NI,NEDDT

COMMON /CURT/ DED,DID,LAMS,BETA

COMMON /CURT/ NCOLS,NSYM,IFER

DDZ=NE(2,J)-PH(1,J)

DNZ=NE(2,J)-PH(1,J)

DNZ=NE(2,J)-NE(1,J)

DNZ=NE(2,J)-NE(1,J)

DNZ=NE(2,J)-NE(1,J)

DNZ=NE(2,J)-NE(1,J)

NRTIE(6,821)

WRITE(6,819)

WRITE(6,819)

WRITE(6,819)

WRITE(6,821)

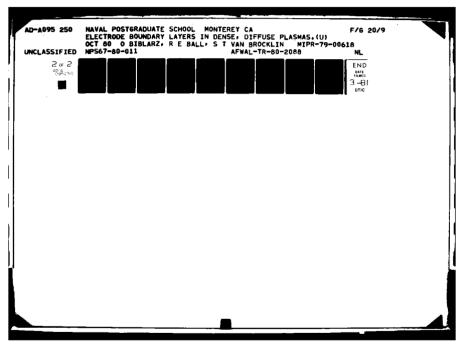
WRITE(6,821)

WRITE(6,821)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          PERPENDICULAR TO WALL AT EACH NODE",/)
                                                                                                                                                   CALCULATES AND PRINTS OUT THE CURRENT TO THE WALL DUE TO ELECTRONS AND IONS
                                           SUBROUTINE JWALL
                                                                                   REDIMENSIONED
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        818
819
820
821
```

- Contract

```
SUBRGUTINE PATCH(Q,XL,XU)
DIMENSION Q(NC,1)
COMMON /IPARM/ NC,NS,IT
DC 100 1=1,NC
DC 100 J=1,NS
IF(Q(I,J).5T.XU) Q(I,J)=XU
IF(Q(I,J).LT.XL) Q(I,J)=XL
ETURN
END
                                                                                                                                    100
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SUBROUTINE WRITEN(Q, COF, LIST)
CCMMON /IPARM/ NC, NS, IT
REAL#4 LIST
DIMENSION Q(NC, 1), T(7), LIST(6)
WRITE(6,801) (NORK, NORK=1, NS)
DO 10 1=1, NC
DO 11 J=1, NC
DO 11 J=1, NC
DO 11 J=1, NC
                       SUBROUTINE WRITEN 800715
                                                                                                                                                                                                 J=1,NS
1(I,J)*COF
16,802) I,(I(J),J=1,NS)
                                                                                                                                                                                                                                800
801
801
802
```

```
SUBROUTINE PUNCHN(Q, COF, LIST)
COMMON / I PARM/ NC, NS, IT
DIMENSION Q(NC, 1), T(7), LIST(6)
REAL#4 LIST
WRITE(7,800) COF, (LIST(KL), KL=1,6)
DC 10 1=1,NC
DO 10 1=1,NC
SWRITE(7,801) I, (T(J), J=1,NS)
OF CRMAT(28, ',644)
FORMAT(28, ',644)
END
                        SUBROUTINE PUNCHN 800715
                                                                                                                                                                   10
800
801
```

) 5 5 5 5 7 7 8 7 TO DISPLAY THE PUNCHED DUTPUT FROM SHEATH IN THREE DIMENSIONAL PRESENTATIONS. DISPLAYS ARE OF THE FORM Z=F(X,Y), WHERE Z IS...
(1) VOLTAGE POTENTIAL DISTRIBUTION
(2) NE DENSITY DISTRIBUTION
(3) NI DENSITY DISTRIBUTION
(4) TEMPERATURE DISTRIBUTION
(5) SPACE CHARGE DISTRIBUTION X(13), Y(19), F(2), SZ(2)) WK(13,19,3) 0), KY(200) 3, 19) IK=I IF(KORK.EQ.2.OR.KORK.EQ.3) IK=NC-I+1 X(I)=FLOAT(I) READ(5.801) (D(IK,J),J=1,NS) XMAX=0.0 EXTERNAL PLOT, PLT301
REAL+8 TTL 121
REAL+4 D(13,7), X(13), Y(13), F(2), SZ(3)
REAL+4 DD(13,19), WK(13,19,3)
INTEGER+4 KX(200), KY(200)
LOGICAL+1 IDN(13,19)
NS=7
FKTR=25
READ(5,802) LAST
READ(5,800) (TTL(KINK), KINK=7,12)
DG 1000 KGRK=11LAST
READ(5,800) (TTL(KINK), KINK=1,6)
NGFQLD=2
JM=NS+NOFQLO-NOFQLO+1
DG 3-11JM
9 Y(J)=FLQAT(J)
DG 10 I=1,NC THREE DIMENSIONAL SHEATH DISPLAY PROGRAM PICT PROGRAM 10

```
ALPHA=45.

BETA=45.

ALPHA=30.

BETA=30.

BETA=30.

BETA=30.

BETA=45.

ALPHA=30.

BETA=45.

BETA=11.

BET
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      BETA RCIATES FIRST ABOUT Z AXIS
ALPHA ROIATES SECOND AROUND Y AXIS
DO 11 1=1.NC

DC 11 J=1.NS

IF(XMAX.LE.D(I.J)) XMAX=D(I.J)

1 CONTINUE

DJ 12 J=1.NS

DC 12 J=1.NS

DC (I.J)=D(I.J)*FKTR

DD(I.S-J)=D(I.J)*FKTR

DD(I.S-J)=D(I.J)*FKTR

DD(I.SO-J)=D(I.J)*FKTR

DD(I.SO-J)=D(I.J)*FKTR

CONTINUE

WRITE(6.808) (NORK,NORK=1.NS)

DC 20 I=1.NS

WRITE(6.809) I,(D(I.J),J=1.NS)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       20
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  000000
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THE ANGDE LINE THE GUTPUT
                                       PROGRAM CONSTRUCTS A PLOT OF
!! POTENTIAL VS. DISTANCE FROM WALL
!) NE AND NI DENSITIES VS. DISTANCE FROM WALL
!) SPACE CHARGE VS. DISTANCE FROM THE WALL
                                                                                                                                                                      K, Y1(K), Y2(K), Y3(K), Y4(K), Y5(K)
                                                                               THE PLOTS ARE OF VALUES TAKEN ALONG OF SYMMETRY. INPUT DATA CONSIST OF FROM THE SHEATH PROGRAM FORMATED AS
                                                                                                                                                       LY(5)/'DIST', ANCE',
LY1(5)/'POTE', NTIA',
LY2(5)/'SPEC', IES',
LY3(5)/'SPAC', ECH',
X(13),Y1(13),Y2(13),
                                                                                                                                                                                                                                                              Y1(1), I=1,NC)
                                                                                                                                                                                                                                                                             Y2(1), [=1,NC)
                                                                                                                                                                                                                                                                                              Y3(I), I=1, NC)
                                                                                                                                                                                                                                                                                                              Y4(I), I=1,NC)
                                                                                                                                                                                                                                                                                                                              (Y5(I), I=1,NC)
                       PROGRAM PLOT
                                                                                                                                                     325
                                        TH
                                                                                                                                                                                                                                                                                                                                             20
```

```
1.5
1.7(NC).0..Y1(1).XL,YL)
1.7(NC).0..1..XL,YL)
1.7(NC).0..1..XL,YL)
1.7(NC).Y5MIN,Y5MAX,XL,YL)
                                                                                                                                                                                                                                                                                                                                                                                    -04*Y1(1)**.
20, LY1,20,0.
20, LY2,20,0.
20, LY3,20,0.
DO 40 I=1NC

IF(YSMIN-6I-YS(I)) YSMIN=YS(I)

CONTINUE

YSMAX=2**YSMAX

YSMAX=2**YSMAX

YSMIN=2**YSMAX

YSMIN=2**YSMAX

YSMIN=2**YSMAX

YSMIN=2**YSMAX

YSMIN=2**YSMAX

YSMIN=2**YSMAX

YSMIN=YCO

CALL PLOTG(X,Y1,NC,1,1,0,LX,20,LC,Y20,LC,Y20,LC,Y20,LC,Y20,LC,Y20,LC,Y20,LC,Y20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LC,Y2,NC,1,1,0,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20,LX,20
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