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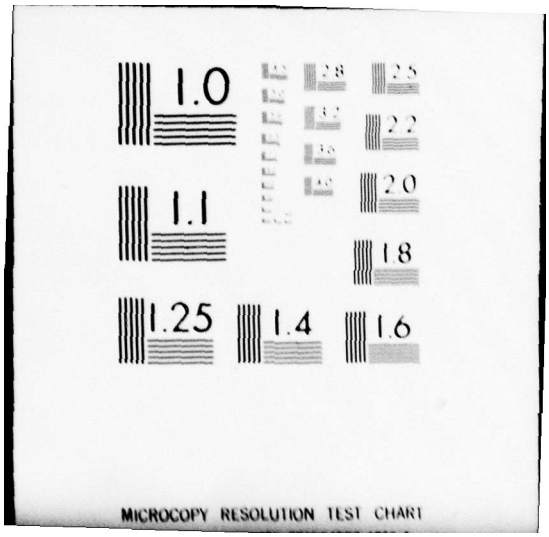
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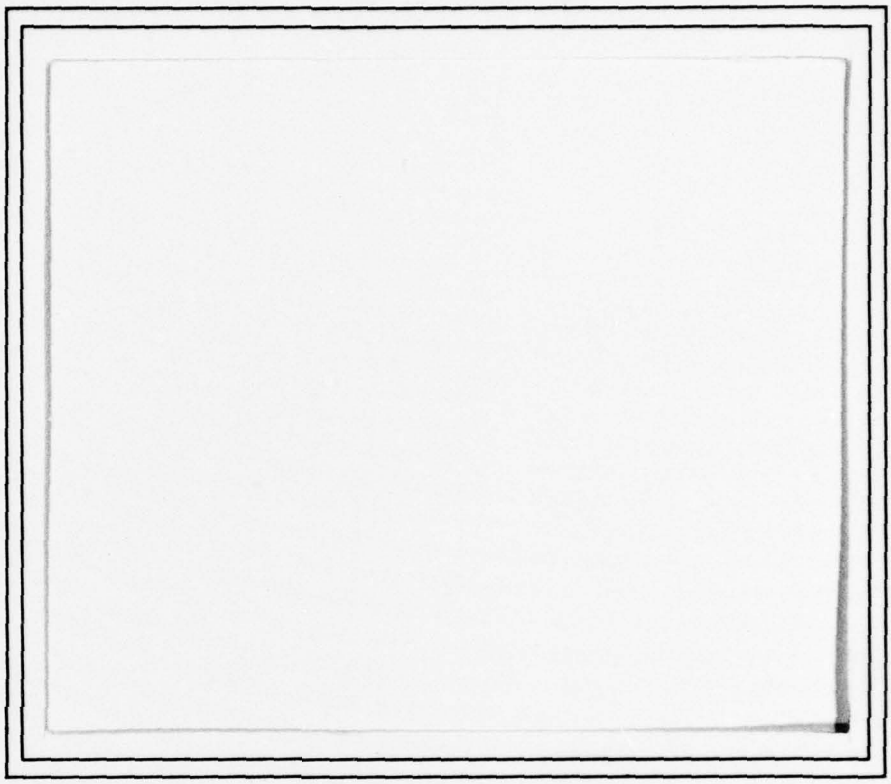

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EDGE DISLOCATION IN NONLOCAL HEXAGONAL ELASTIC CRYSTALS*

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ABSTRACT

The solution is presented for the problems of edge dislocation in hexagonal crystals with long range interatomic interactions. The field equations of nonlocal elastic solids are employed to determine the stress fields and the elastic energy for an edge dislocation in the basal plane. Classical stress and energy singularities are found not present in the nonlocal solutions. Stress distribution is calculated and maximum shear stress is given for various hexagonal materials. Theoretical shear stress to initiate a dislocation having a Burger's vector of one atomic distance is calculated and found to be in the acceptable range known from the lattice dynamic calculations.

1. INTRODUCTION

It is well-known that the classical elasticity solution of the edge dislocation contains stress and energy singularities in the "core region", cf. [1]. In several previous papers (e.g., [2], [3], [4]) we have shown that the solutions of various Volterra dislocations based on the nonlocal elasticity theory do not contain

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these singularities. This recent theory [5,6] models the elastic materials much more satisfactorily in that the effect of long range interatomic interactions are taken into account. It seems that no artifice such as introducing various atomistic models to estimate the stress and energy in the core region is necessary. Moreover, as a continuum theory all problems can be reduced to boundary-initial value problems.

The discussion of the dislocation problems in anisotropic solids is not a trivial extension even in its classical frame of reference. Moreover because of the orientational effects the state of stress and elastic energy are affected considerably. Consequently the criteria for failure or the generation of dislocations need new investigations. The raison d'ętre of the present paper stems from these considerations. In Section 2 we present a brief summary of the field equations of the nonlocal elasticity theory. In Section 3 we obtain the solution of the edge dislocation problem leading to stress and energy fields and in Section 4 we specialize these results to the isotropic crystals. In Section 5, some results of computer calculations are presented and maximum stresses that cause a single edge dislocation in several hexagonal crystals (Mg. Apatite, Cd, Zn) are calculated. The distribution of normal and shear stresses along radial line r, $\theta=0$, and as a function of the polar angle θ for a fixed r are calculated. Since no stress and energy singularity occur, the maximum stress hypothesis may be used to calculate the theoretical (cohesive) stress. The results are in the accepted range known from the atomic theory.

2. FORMULATION

In previous papers [2,4] we have shown that, under some very general conditions, the solution of elastostatic problems in linear nonlocal elasticity can be reduced to the solution of the classical Navier's equation, however the stress field is calculated by

$$t_{k\ell}(\underset{\sim}{x}) = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\sigma_{k\ell}(\underset{\sim}{x}')dv(\underset{\sim}{x}') \tag{2.1}$$

where $\sigma_{k\ell}$ is given by the classical Hooke's law which for the hexagonal crystals can be arranged into the form, cf. Fig. 1a,

$$\begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{23} \\ \sigma_{31} \\ \sigma_{12} \end{bmatrix} = \begin{bmatrix} C_{11} & C_{12} & C_{13} & & & \\ C_{12} & C_{22} & C_{12} & & & \\ C_{13} & C_{12} & C_{11} & & & \\ & & & 2C_{44} & & \\ & & & & 2C_{55} & \\ & & & & & 2C_{44} \end{bmatrix} \begin{bmatrix} e_{11} \\ e_{22} \\ e_{33} \\ e_{23} \\ e_{31} \\ e_{12} \end{bmatrix} . \tag{2.2}$$

Here $e_{k\ell}$ is the linear strain tensor given by

$$e_{k\ell} \equiv \frac{1}{2}(u_{k,\ell} + u_{\ell,k}) \tag{2.3}$$

where $u_{k,\ell} \equiv \partial u_k/\partial x_\ell$. Since for the hexagonal crystals $2C_{55}=C_{11}-C_{13}$, the number of independent elastic constants are five.

The attenuation function $\alpha(\underset{\sim}{x}'-\underset{\sim}{x})$ suggested in our previous work is of the form

$$(2.4) \qquad \alpha = \alpha_o \exp[-(k_1/a)^2(x'_\beta-x_\beta)(x'_\beta-x_\beta)-(k_2/a)^2(x'_2-x_2)^2]$$

where β is summed over $\beta=1$ and $\beta=3$. Here a is the lattice parameter k_1 and k_2 are two constants which govern the range of attenuation of the interatomic attractions. The constant α_o is determined from the normalization condition on α:

$$(2.5) \qquad \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\, dv(\underset{\sim}{x}') = 1$$

so that

$$(2.6) \qquad \alpha_o = k_1^2 k_2/\pi^{3/2}\, a^3 \quad .$$

Now a problem in nonlocal elasticity is reduced to determining the displacement field $u_k(\underset{\sim}{x})$ by solving the Navier's equation obtained by combining (2.2) and (2.3) with the Cauchy's equation

$$(2.7) \qquad \sigma_{k\ell,k} = 0 \quad , \quad k,\ell = 1,2,3$$

where as usual repeated indices are summed over (1,2,3). Once $u_k(\underset{\sim}{x})$ is determined one can then calculate strain from (2.3) and the stress field $t_{k\ell}$ by using (2.2) in (2.1). We now apply this program to solve the problem of the edge dislocations.

3. EDGE DISLOCATION

The straight edge dislocation in a hexagonal crystal is possible in the basal plane y=const. in the x-direction Fig. 1a. We vision such a dislocation by cutting a cylinder along a radial plane and pulling lower surface by a constant amount b (called Burger's vector) relative to the upper surface and welding the two surfaces, Fig. 1b. The solution of this problem in classical elasticity is well-known (cf. [1], p. 422). The displacement field $(u_x,u_y,0)$ satisfying Navier's equation, is given by

$$(3.1)\qquad u_x = \frac{b}{4\pi}\left[\arctan\left(\frac{u_1 xy}{x^2-\lambda^2 y^2}\right) + u_2 \ell n(q/t)\right],$$

$$u_y = \frac{-b}{4\pi}\, u_3\left[u_4 \ell n(qt) - u_5 \arctan\left(\frac{u_6 y^2}{x^2-u_7 y^2}\right)\right]$$

and $\sigma_{k\ell}$ by

$$(3.2)\qquad \sigma_{xx} = -\frac{Mb}{2\pi}\,\frac{\sigma_1 x^2 y+\sigma_2 y^3}{x^4+p_1 x^2 y^2+p_2 y^4},$$

$$\sigma_{yy} = \frac{Mb}{2\pi}\,\frac{\sigma_3 x^2 y-\sigma_4 y^3}{x^4+p_1 x^2 y^2+p_2 y^4},$$

$$\sigma_{xy} = \frac{Mb}{2\pi}\,\frac{\sigma_3 x^3-\sigma_4 xy^2}{x^4+p_1 x^2 y^2+p_2 y^4}$$

all other components of $\sigma_{k\ell}$ and u_z vanish. Here we used (x,y,z) for subscripts $(1,2,3)$ and set

(3.3) $\bar{C}_{11} \equiv (C_{11}C_{22})^{1/2}$,

$M \equiv C_{44}(\bar{C}_{11}+C_{12})\{(\bar{C}_{11}-C_{12})/[C_{22}C_{44}(\bar{C}_{11}+C_{12}+2C_{44})]\}^{1/2}$,

$\sigma_1 \equiv [(\bar{C}_{11}-C_{12})(\bar{C}_{11}+C_{12}+2C_{44})-\bar{C}_{11}C_{44}]/(C_{22}C_{44})$,

$\sigma_2 \equiv C_{11}/C_{22}$, $\sigma_3 \equiv 1$, $\sigma_4 \equiv (C_{11}/C_{22})^{1/2}$,

$p_1 = 2(\bar{C}_{11}/C_{22})+(\bar{C}_{11}+C_{12})(\bar{C}_{11}-C_{12}-2C_{44})/(C_{22}C_{44})$,

$p_2 \equiv C_{11}/C_{22}$,

$\lambda \equiv (C_{11}/C_{22})^{1/4}$, $\phi \equiv \frac{1}{2}\arccos[(C_{12}^2+2C_{12}C_{44}-\bar{C}_{11}^2)/(2\bar{C}_{11}C_{44})]$,

$q^2 \equiv x^2+2\lambda\cos\phi xy+\lambda^2y^2$ $t^2 \equiv x^2-2\lambda\cos\phi xy+\lambda^2y^2$,

$u_1 \equiv 2\lambda\sin\phi$, $u_2 \equiv (\bar{C}_{11}^2-C_{12}^2)/(2\bar{C}_{11}C_{44}\sin2\phi)$,

$u_3 \equiv \lambda/(\bar{C}_{11}\sin2\phi)$, $u_4 \equiv (\bar{C}_{11}-C_{12})\cos\phi$,

$u_5 \equiv (\bar{C}_{11}+C_{12})\sin\phi$, $u_6 \equiv \lambda^2\sin2\phi$, $u_7 \equiv \lambda^2\cos2\phi$

Introducing the cylindrical coordinates by (Fig. 2),

(3.4) $x = r\cos\theta$, $y = r\sin\theta$, $z = z$

$x' = r'\cos\theta'$, $y' = r'\sin\theta'$, $z' = z'$

$dv(\underset{\sim}{x}') = r'\,dr'\,d\theta'\,dz'$

we have for the components of $\sigma_{k\ell}$ in cylindrical coordinates:

$$(3.5)\qquad \sigma_{rr} = \frac{\sigma_{xx}+\sigma_{yy}}{2} + \frac{\sigma_{xx}-\sigma_{yy}}{2}\cos 2\theta + \sigma_{xy}\sin 2\theta \quad ,$$

$$\sigma_{\theta\theta} = \frac{\sigma_{xx}+\sigma_{yy}}{2} - \frac{\sigma_{xx}-\sigma_{yy}}{2}\cos 2\theta - \sigma_{xy}\sin 2\theta \quad ,$$

$$\sigma_{r\theta} = -\frac{\sigma_{xx}-\sigma_{yy}}{2}\sin 2\theta + \sigma_{xy}\cos 2\theta \quad .$$

The physical components $t^{(k)}(\ell)$ of the stress tensor in the nonlocal theory are given in terms of those of the classical stresses $\sigma^{(k)}(\ell)$ and shifters $\delta^{k}{}_{k'}$ by

$$(3.6)\qquad t^{(k)}(\ell) = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\sigma^{(k')}(\ell')\delta^{\ell'}{}_{\ell}\,\delta^{k}{}_{k'}\,dv(\underset{\sim}{x}')$$

cf. [2]. In cylindrical coordinates the shifters have the following values

$$(3.7)\qquad \delta^{1'}{}_{1} = \delta^{2'}{}_{2} = \underset{\sim}{e}_r\cdot\underset{\sim}{e}'_r = \underset{\sim}{e}_\theta\cdot\underset{\sim}{e}'_\theta = \cos(\theta'-\theta) \quad ,$$

$$\delta^{1'}{}_{2} = \underset{\sim}{e}'_r\cdot\underset{\sim}{e}_\theta = \sin(\theta'-\theta) \quad , \quad \delta^{1}{}_{2'} = \underset{\sim}{e}_r\cdot\underset{\sim}{e}'_\theta = -\sin(\theta'-\theta)$$

$$\delta^{3}{}_{3'} = \underset{\sim}{e}_z\cdot\underset{\sim}{e}'_z = 1 \quad , \quad \delta^{k}{}_{k'} = \delta^{k'}{}_{k} \quad .$$

Thus the physical components of the stress tensor have the form

$$(3.8)\quad t_{rr} = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\left[\frac{\sigma'_{rr}+\sigma'_{\theta\theta}}{2} + \frac{\sigma'_{rr}-\sigma'_{\theta\theta}}{2}\cos(\theta'-\theta) - \sigma'_{r\theta}\sin 2(\theta'-\theta)\right]dv(\underset{\sim}{x}') \quad ,$$

$$t_{\theta\theta} = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\left[\frac{\sigma'_{rr}+\sigma'_{\theta\theta}}{2} - \frac{\sigma'_{rr}-\sigma'_{\theta\theta}}{2}\cos(\theta'-\theta) + \sigma'_{r\theta}\sin 2(\theta'-\theta)\right]dv(\underset{\sim}{x}') \quad ,$$

$$t_{r\theta} = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\left[\frac{\sigma'_{rr}-\sigma'_{\theta\theta}}{2}\sin 2(\theta'-\theta) + \sigma'_{r\theta}\cos 2(\theta'-\theta)\right]dv(\underset{\sim}{x}') \quad ,$$

where we used the abreviation $\sigma'_{ij} \equiv \sigma_{ij}(\underset{\sim}{x}')$. Using the relations

$$(3.9)\quad \frac{\sigma'_{rr}+\sigma'_{\theta\theta}}{2} = \frac{\sigma'_{xx}+\sigma'_{yy}}{2} \quad , \quad \frac{\sigma'_{rr}-\sigma'_{\theta\theta}}{2} = \frac{\sigma'_{xx}-\sigma'_{yy}}{2}\cos 2\theta' + \sigma'_{xy}\sin 2\theta'$$

(3.8) may be written as

$$(3.10)\quad t_{rr} = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\left[\frac{\sigma'_{xx}+\sigma'_{yy}}{2} + \frac{\sigma'_{xx}-\sigma'_{yy}}{2}\cos 2\theta + \sigma'_{xy}\sin 2\theta\right]dv(\underset{\sim}{x}') \quad ,$$

$$t_{\theta\theta} = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\left[\frac{\sigma'_{xx}+\sigma'_{yy}}{2} - \frac{\sigma'_{xx}-\sigma'_{yy}}{2}\cos 2\theta - \sigma'_{xy}\sin 2\theta\right]dv(\underset{\sim}{x}') \quad ,$$

$$t_{r\theta} = \int_V \alpha(\underset{\sim}{x}'-\underset{\sim}{x})\left[-\frac{\sigma'_{xx}-\sigma'_{yy}}{2}\sin 2\theta + \sigma'_{xy}\cos 2\theta\right]dv(\underset{\sim}{x}') \quad .$$

The attenuation function $\alpha(\underset{\sim}{x}'-\underset{\sim}{x})$ in cylindrical coordinates acquires the form

$$(3.11)\quad \alpha(\underset{\sim}{x}'-\underset{\sim}{x}) = \pi^{-3/2}(k_1^2 k_2/a^3)\exp[-(k_1^2/a^2)(z'-z)^2]$$

$$\cdot \exp[-(r^2/a^2)(k_1^2\cos^2\theta + k_2^2\sin^2\theta)]$$

$$\cdot \exp[-(k_1^2/a^2)(r'^2\cos^2\theta' - 2rr'\cos\theta\cos\theta')$$

$$-(k_2^2/a^2)(r'^2\sin^2\theta' - 2rr'\sin\theta\sin\theta')] \quad .$$

We now substitute (3.4) into (3.2), the result and (3.11) into (3.10). This, for each of the stress components, leads to a triple integral over the domain ($0\leq r'<\infty$, $0\leq\theta'<2\pi$, $-\infty<z'<\infty$). The integrations over r' and z' are tedious but can be carried out, however the integration over θ' will have to be done numerically. Leaving the details of these calculations we give the results:

$$(3.12)\qquad \{t_{rr},t_{\theta\theta},t_{r\theta}\} = \int_0^{2\pi} \{T_{rr}(\theta,\theta'),\ T_{\theta\theta}(\theta,\theta'),\ T_{r\theta}(\theta,\theta')\}\cdot f(r,\theta,\theta')d\theta'$$

where

$$(3.13)\qquad T_{rr} = -t_o\,(g_1 + g_2\cos2\theta - g_3\sin2\theta)$$

$$T_{\theta\theta} = -t_o\,(g_1 - g_2\cos2\theta + g_3\sin2\theta)$$

$$T_{r\theta} = t_o\,(g_2\sin2\theta + g_3\cos2\theta)$$

$$g_1 = q_1\cos^2\theta'\sin\theta' + q_2\sin^3\theta' \ ,$$

$$g_2 = q_3\cos^2\theta'\sin\theta' + q_4\sin^3\theta' \ ,$$

$$g_3 = 2(\sigma_3\cos^3\theta' - \sigma_4\cos\theta'\sin^2\theta') \ ,$$

$$f(r,\theta,\theta') = \frac{\kappa}{4\pi^{1/2}} \exp\{-(kr/a)^2[1+(\kappa^2-1)\sin^2\theta - \frac{(\cos\theta\cos\theta'+\kappa^2\sin\theta\sin\theta')^2}{1+(\kappa^2-1)\sin^2\theta'}]\}$$

$$\cdot\{1+\mathrm{erf}[(kr/a)\frac{\cos\theta\cos\theta'+\kappa^2\sin\theta\sin\theta'}{[1+(\kappa^2-1)\sin^2\theta']^{1/2}}]\}$$

$$\cdot(\cos^4\theta'+p_1\sin^2\theta'\cos^2\theta'+p_2\sin^4\theta')^{-1}[1+(\kappa^2-1)\sin^2\theta']^{-1/2}$$

where we also set

$$(3.14)\qquad q_1 \equiv \sigma_1-\sigma_3 \ , \ q_2 \equiv \sigma_2+\sigma_4 \ , \ q_3 \equiv \sigma_1+\sigma_3 \ , \ q_4 \equiv \sigma_2-\sigma_4 \ ,$$

$$k_1 \equiv k \ , \ k_2/k_1 \equiv \kappa \ , \ t_o \equiv Mbk/2\pi a \ .$$

The total strain energy is given by

$$(3.15)\qquad \Sigma = 2L \int_0^R\int_0^{\pi/2} (t_{rr}e_{rr}+t_{\theta\theta}e_{\theta\theta}+2t_{r\theta}e_{r\theta})r\ dr\ d\theta$$

where L is the length of the cylinder and R is the radius. Substituting for e_{rr}, $e_{\theta\theta}$ and $e_{r\theta}$ calculated from (2.3), in the same way as in the case for σ_{rr},$\sigma_{\theta\theta}$, $\sigma_{r\theta}$, we write

$$(3.16)\qquad \Sigma = \frac{MbL}{\pi}\int_0^R\int_0^{\pi/2} [t_{rr}(a_1\cos^4\theta+a_2\cos^2\theta+a_3)\sin\theta+t_{\theta\theta}(a_4\cos^4\theta$$

$$+a_5\cos^2\theta+a_6)\sin\theta+t_{r\theta}(a_7\sin^4\theta+a_8\sin^2\theta+a_9)$$

$$\cos\theta](\cos^4\theta+p_1\sin^2\theta\cos^2\theta+p_2\sin^4\theta)^{-1}d\theta dr$$

where

(3.17) $$a_1 \equiv (\varepsilon_2-\varepsilon_1)(\sigma_1-\sigma_2)+(\varepsilon_2-\varepsilon_3+\varepsilon_4)(\sigma_3+\sigma_4) \quad ,$$

$$a_2 \equiv -\varepsilon_2\sigma_1+(2\varepsilon_2-\varepsilon_1)\sigma_2+\varepsilon_3\sigma_3+(2\varepsilon_3-\varepsilon_2-\varepsilon_4)\sigma_4 \quad ,$$

$$a_3 \equiv -(\varepsilon_2\sigma_2+\varepsilon_3\sigma_4) \quad , \quad a_4 \equiv -a_1 \quad ,$$

$$a_5 \equiv -\varepsilon_1\sigma_1+(2\varepsilon_1-\varepsilon_2)\sigma_2+\varepsilon_2\sigma_3+(2\varepsilon_2-\varepsilon_3+\varepsilon_4)\sigma_4 \quad ,$$

$$a_6 \equiv -(\varepsilon_1\sigma_2+\varepsilon_2\sigma_4) \quad , \quad a_7 \equiv 2a_1 \quad ,$$

$$a_8 \equiv 2[(\varepsilon_1-\varepsilon_2)\sigma_1-(\varepsilon_2-\varepsilon_3+\varepsilon_4)\sigma_3]-\varepsilon_4(\sigma_3+\sigma_4) \quad ,$$

$$a_9 \equiv \varepsilon_4\sigma_3 \quad ,$$

and

$$(\varepsilon_1,\varepsilon_2,\varepsilon_3) \equiv (C_{22},-C_{12},C_{11})(C_{11}C_{22}-C_{12}^2)^{-1} \quad , \quad \varepsilon_4 \equiv C_{44}^{-1} \quad .$$

4. ISOTROPIC SOLID

In the case of isotropic solids we have

(4.1) $$M = \mu/(1-\nu) \quad , \quad \kappa=1 \quad , \quad q_1=q_2=2 \quad , \quad q_3=4 \quad , \quad q_4=0 \quad ,$$

$$\sigma_1=3 \quad , \quad \sigma_2=\sigma_3=\sigma_4=1 \quad , \quad p_1=2 \quad , \quad p_2=1 \quad ,$$

$$a_1=a_2=a_4=a_5=a_7=a_8=0 \quad ,$$

$$a_3=a_6= -[2(\lambda+\mu)]^{-1} \quad , \quad a_9=1/\mu \quad .$$

Using these values in the expressions of (3.12), (3.13), (3.15) and (3.16) we obtain

$$(4.2) \qquad t_{rr} = - \frac{\mu bk}{2\pi(1-\nu)a} \{1-[1-\exp(-p^2)]p^{-2}\}p^{-1} \sin\theta \quad ,$$

$$t_{\theta\theta} = \frac{\mu bk}{2\pi(1-\nu)a} \{1-(2+p^{-2})[1-\exp(-p^2)]\}p^{-1} \sin\theta \quad ,$$

$$t_{r\theta} = \frac{\mu bk}{2\pi(1-\nu)a} \{1-[1-\exp(-p^2)]p^{-2}\}p^{-1} \cos\theta \quad ,$$

$$(4.3) \qquad \Sigma = \frac{\mu b^2 L}{16\pi(1-\nu)^2} \{2(1-\nu)[C+\ell np^2+E_1(p^2)]-1+[1-\exp(-p^2)]p^{-2}\} \quad ,$$

where C is the Euler's constant and

$$(4.4) \qquad p \equiv kr/a \quad , \quad E_1(x) \equiv \int_x^\infty \frac{e^{-t}}{t} dt \quad , \quad C = 0.5772... \quad .$$

These results are in complete agreement with those given in [3].

5. DISCUSSION

Components of the stress tensor are plotted in Fig. 3 as a function of the polar angle θ for a fixed radial distance. It is observed that t_{rr} and $t_{\theta\theta}$ have the same shape with their extrema occurring at the same angle. The extremum values of $t_{r\theta}$ differ by an angle $\pi/2$ from those of the normal stresses. For the fracture calculations the state of stress at $\theta=0$ and $\pi/2$ are important. It is known that the cleavage stress

of the crystals is at least twice the maximum shear stress (Kelly [7], p. 17). Of the two state of stress investigated it is found that the one at $\theta=0$ plane is the most important. In Fig. 4 the shear stress is plotted as a function of $p=kr/a$ for a few hexagonal crystals (Zn, Cd, Apatite, Mg). The elastic constants of these materials listed in Table 1 are taken from ref. [8]. The shear stress for an isotropic crystal (with $\nu=0.3$) is also plotted in Fig. 4. The ratio of the shear stress in hexagonal crystals to that of the isotropic solids may be useful from the point of view of technological applications. This is given in Fig. 5 for the same crystals. Finally we give a plot of the elastic energy as a function of the radius of the cylinder, Fig. 6. Of course as the radius of the cylinder R increases the elastic energy also increases. We note however that no singularity is present either in the stress field or in the energy. The usual singularities present in the classical elasticity solutions however appear when the lattice parameter $a\to 0$. This is the classical continuum limit. In Table 1 we give the material moduli, the maximum shear stress and its radial location. The ratio of the attenuation constants κ is taken, by an anology to the wave propagation [10, ch. 6], as $\kappa=2C_{22}/(C_{11}+C_{22})$. In Table 2 the energy ratio of the hexagonal crystals to the isotropic solids is listed for various radii of the cylindrical specimens. Finally, the maximum value of the ratio $t_{r\theta}/C_{44}$ may be used to estimate the theoretical shear strength of the crystal. To compute this we need to estimate the attenuation constant k in $\alpha(\underset{\sim}{x}'-\underset{\sim}{x})$. This function decreases to its one percent value at n lattice parameter distance if

$$k = 2.146/n \quad .$$

As an example, for zinc we have

$$(kb/a)^{-1}\, t_{r\theta\ max} = 0.197\ 10^{11}\ \text{dyn/cm}^2 \quad .$$

If we choose b=a ([9], p. 516) then

$$t_{r\theta\ max}/C_{44} = \frac{0.197\ 10^{11}}{3.96\ 10^{11}}\ k = 0.050k \quad .$$

The maximum value that Kelly ([7], p. 19) gives

$$t_{r\theta\ max}/C_{44} = 0.034 \quad .$$

Thus by comparing we obtain

$$k = 0.034/0.050 = 0.68 \quad , \quad n = 2.146/0.68 = 3.14 \quad .$$

The maximum values of $t_{r\theta}/C_{44}$ are shown on Table 3 for various hexagonal crystals for n=2, n=2.5 and n=3.0. These results are in the right range as predicted by other methods and atomic considerations. For example, in the atomic theory of crystal lattices it is known that one must take into account the interactions of at least eight closest neighbours to obtain a result consistent with experiments [11, 12]. For hexagonal crystals eighth neighbours corresponds to n=2.

While clearly there is one parameter (namely k) which need be estimated, the range of this parameter can be ascertained from our knowledge in condensed matter. The detail shape of the nonlocal

moduli seems to be less effective so long as it is a candidate to be a distribution. The flexibility in the choice of $\alpha(x'-x)$ and k should be considered an asset in the sense that for noncrystalline materials and imperfect crystals the attenuation function can be adjusted for a given material once and for all. Afterward all problems for such a solid are reduced to boundary-value problems.

REFERENCES

[1] Hirth, J.P. and Lothe, J. [1968]: Theory of Dislocations, McGraw-Hill.

[2] Eringen, A.C. [1977]: "Screw Dislocation in Nonlocal Elasticity", J. Phys. D: Appl. Phys., Vol. 10, pp. 671-678.

[3] Eringen, A.C. [1977]: "Edge Dislocation in Nonlocal Elasticity", Int. J. Engng. Sci., Vol. 15, pp. 177-183.

[4] Eringen, A.C. and F. Balta [1978]: "Screw Dislocation in Nonlocal Hexagonal Elastic Crystals", scheduled for publication in Crystal Lattice Defects.

[5] Eringen, A.C. [1972]: "Linear Theory of Nonlocal Elasticity and Dispersion of Plane Waves", Int. J. Engng. Sci., 10, 425-435.

[6] Eringen, A.C. [1976]: Continuum Physics, Vol. IV, Acad. Press.

[7] Kelly, A. [1966]: Strong Solids, Oxford.

[8] Hearmon, R.F.S. [1966, 1969]: "The Elastic Constants of Non-Piezoelectric Crystals", Landolf-Börnstein, Numerical Data and Functional Relationships in Science, ed. by Hellwage, K.H., Vols. III/1 and III/2, Springer-Verlag.

[9] Dorn, J.E. and Mitchell, J.B., [1965]: "Slip Mechanism in Single Crystals of Hexagonal Close-Packed Phases", High Strength Materials, ed. by Zackay, V.F., Wiley.

[10] Federov, F.I. [1968]: Elastic Waves in Crystals, Plenum.

[11] Upadhyaya, J.C. and Verma, M.P. [1973]: "Dispersion Relations in some Hexagonal Metals", J. Phys. F: Metal Phys., Vol. 3, p. 640.

[12] Bertoni, C.M., et.al. [1975]: "Three Body Forces in hcp Metals", J. Phys. F: Metal Phys. Vol. 5, p. 419.

Table 1. Maximum Shear Stresses

Material	Elastic Constants $\times 10^{11}$ dyn/cm^2					κ	$t_{r\theta}/t_o$	F_m
	C_{11}	C_{12}	C_{13}	C_{22}	C_{44}			
Zn	16.5	5.0	3.1	6.2	3.96	0.546	0.217	2.30
Mg	5.93	2.14	2.57	6.15	1.64	1.02	0.398	1.45
Cd	11.4	4.00	3.94	5.08	2.00	0.617	0.237	2.05
Apatite	16.7	6.6	1.31	14.0	6.63	0.912	0.389	1.55
Ice(257K)	1.34	0.53	0.65	1.45	0.313	1.04	0.396	1.45

Table 2. Strain Energy Ratio Σ/Σ_o

Material	p=kr/a				
	1.	2.	5.	10.	20.
Isotropic(ν=.3)	0.747	2.000	4.355	6.265	8.199
Isotropic(κ=1 ν=.3)	0.748	2.001	4.354	6.263	8.195
Zn	0.265	0.889	2.751	4.563	6.462
Mg	0.739	1.956	4.204	6.021	7.858
Cd	0.260	0.846	2.456	3.960	5.522
Apatite	0.845	2.338	5.272	7.679	10.121
Ice	0.675	1.777	3.807	5.448	7.107

Table 3. Shear Stress $t_{r\theta}/C_{44}$

Material	n		
	2.0	2.5	3.0
Zn	0.053	0.043	0.036
Mg	0.102	0.082	0.068
Cd	0.071	0.057	0.047
Apatite	0.075	0.060	0.050
Ice	0.112	0.090	0.075

FIGURE CAPTIONS

Figure 1: (a) Hexagonal crystal
(b) Straight edge dislocation

Figure 2: Cylindrical coordinates

Figure 3: Stresses in xy plane versus θ, p=2.30.

Figure 4: $t_{r\theta}/t_o$ versus p, θ=0. (ν=0.3 for isotropic case)

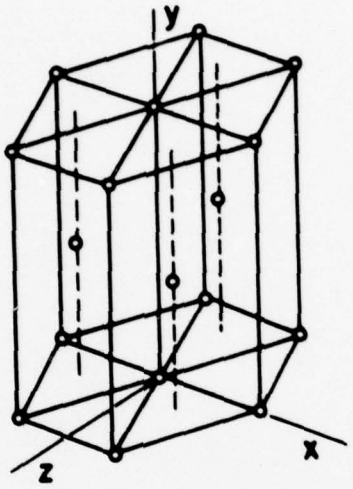
Figure 5: $t_{r\theta}/t^o_{r\theta}$, ratio of anisotropic to isotropic shear stress at θ=0. (ν=0.3 for isotropic case)

Figure 6: Σ/Lb^2 versus p=kR/a

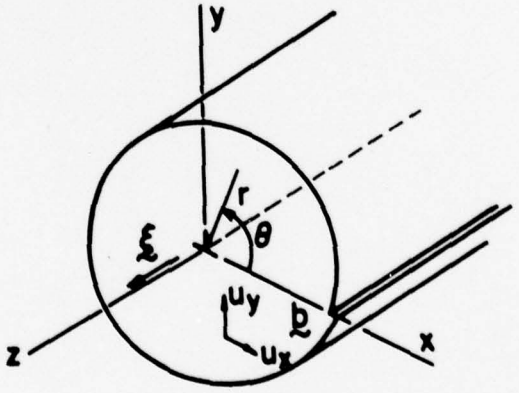
Table 1: Elastic constants and maximum shear stresses

Table 2: Strain energy ratio Σ/Σ_o

Table 3: Shear stress $t_{r\theta}/C_{44}$



(a)



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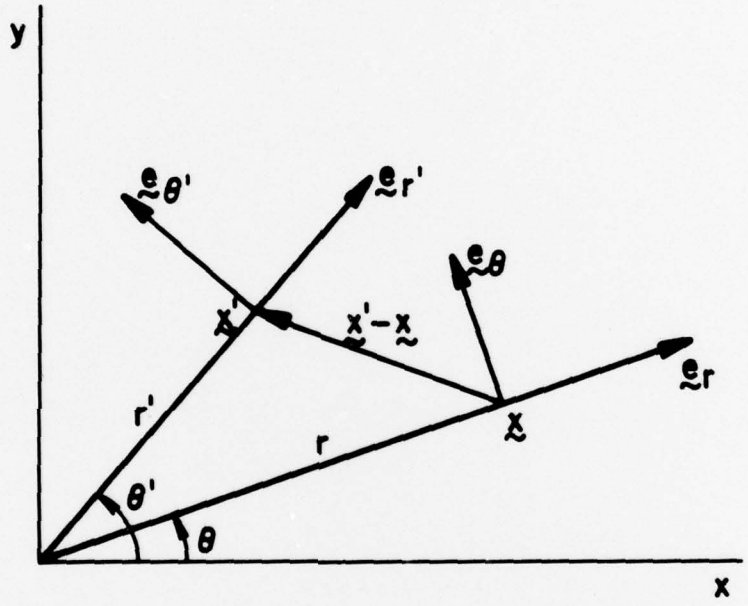
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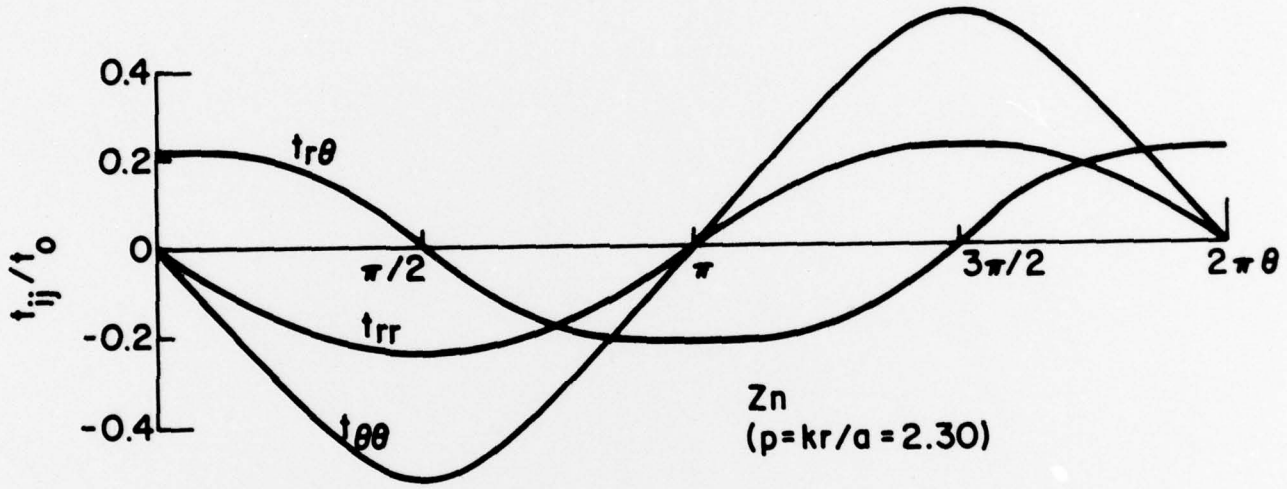
$\underset{\sim}{e}_{\theta'}$
$\underset{\sim}{e}_{r'}$
$\underset{\sim}{e}_{\theta}$
$\underset{\sim}{x}'$
$\underset{\sim}{x}'-\underset{\sim}{x}$
$\underset{\sim}{e}_{r}$
r'
r
$\underset{\sim}{x}$
θ'
θ
x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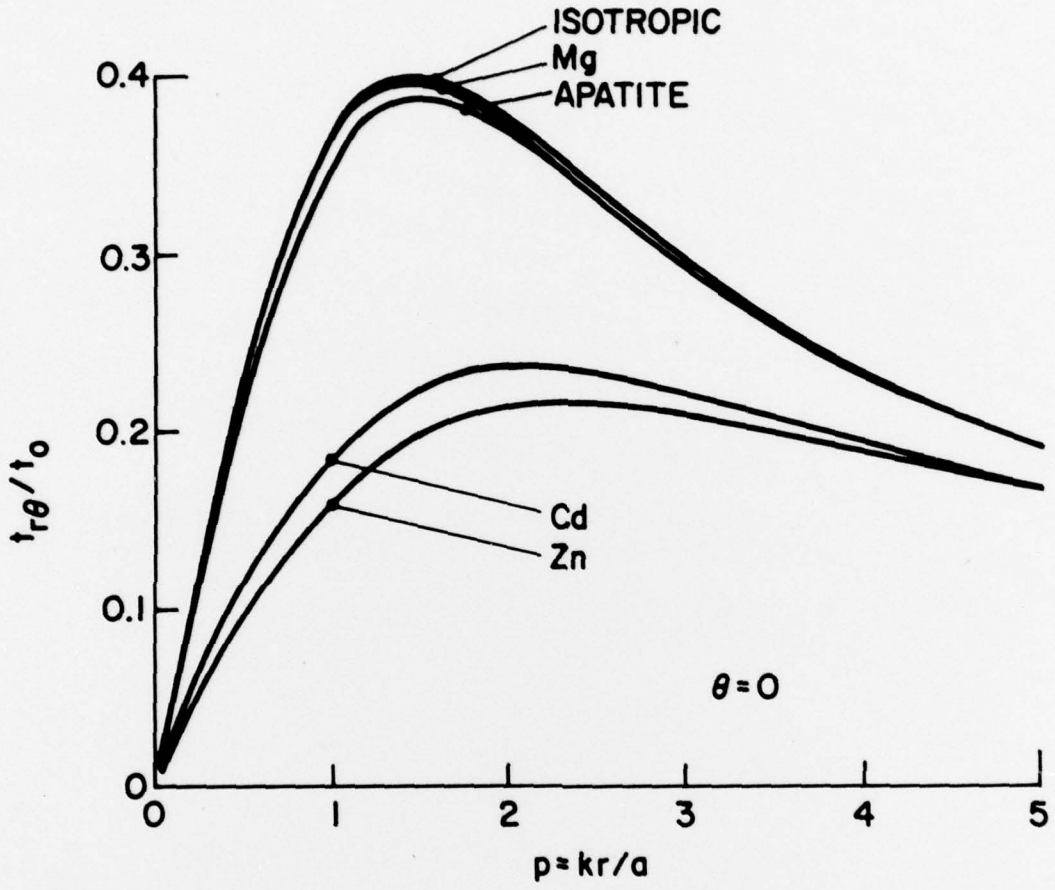
0.2
0
-0.2
-0.4
t_{ij}/t_o
$t_{r\theta}$
t_{rr}
$t_{\theta\theta}$
$\pi/2$
π
$3\pi/2$
$2\pi\theta$
Zn
(p=kr/a=2.30)


ISOTROPIC
Mg
APATITE
Cd
Zn
θ=0
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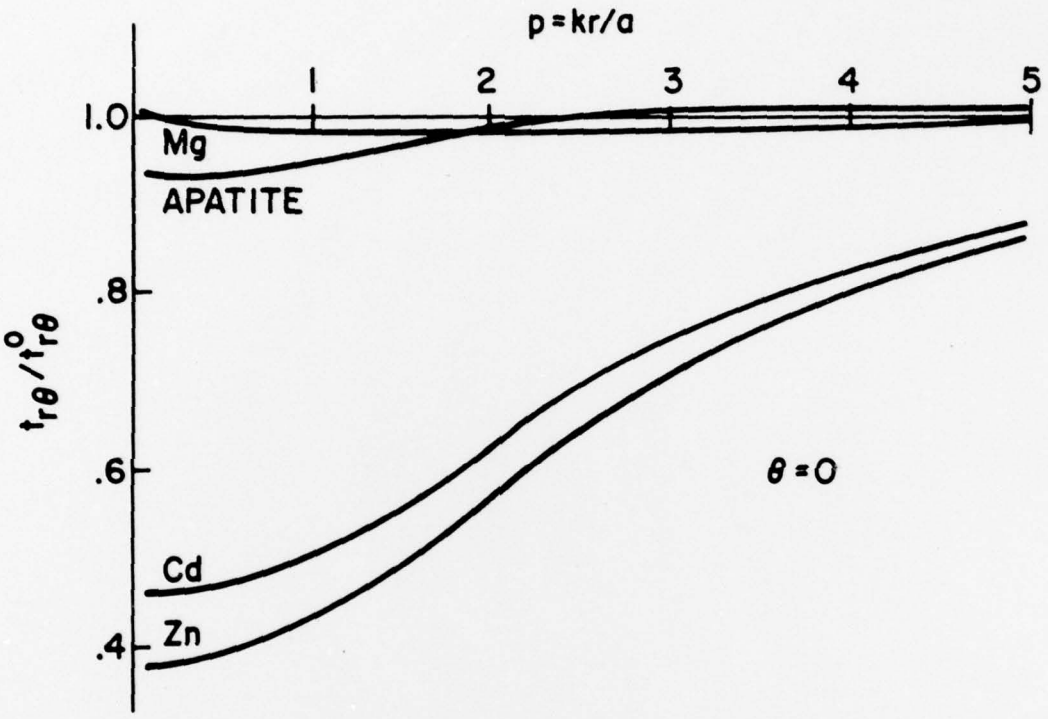
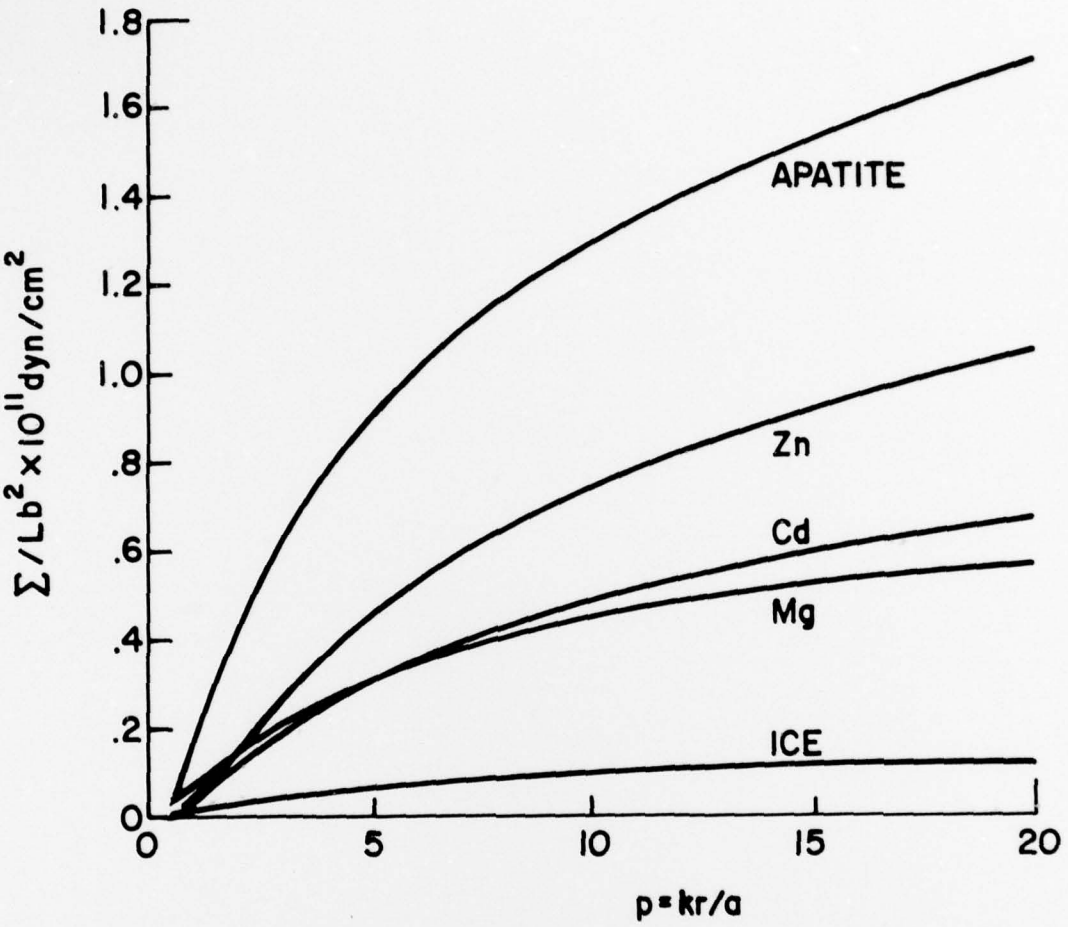
0.2
0.1
0
0
1
2
3
4
5
$t_{r\theta}/t_o$
p=kr/a


p=kr/a
1
2
3
4
5
1.0
.8
.6
.4
Mg
APATITE
Cd
Zn
θ=0
$t_{r\theta}/t^{o}_{r\theta}$


1.8
1.6
1.4
1.2
1.0
.8
.6
.4
.2
0
$\Sigma/Lb^2 \times 10^{11}$ dyn/cm^2
APATITE
Zn
Cd
Mg
ICE
0
5
10
15
20
p = kr/a

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The solution is presented for the problems of edge dislocation in hexagonal crystals with long range interatomic interactions. The field equations of nonlocal elastic solids are employed to determine the stress fields and the elastic energy for an edge dislocation in the basal plane. Classical stress and energy singularities are found not present in the nonlocal solutions. Stress distribution is calculated and maximum shear stress is given for various hexagonal materials. Theoretical shear stress to initiate a dislocation having a Burger's vector of one atomic distance is calculated and found to be in the

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