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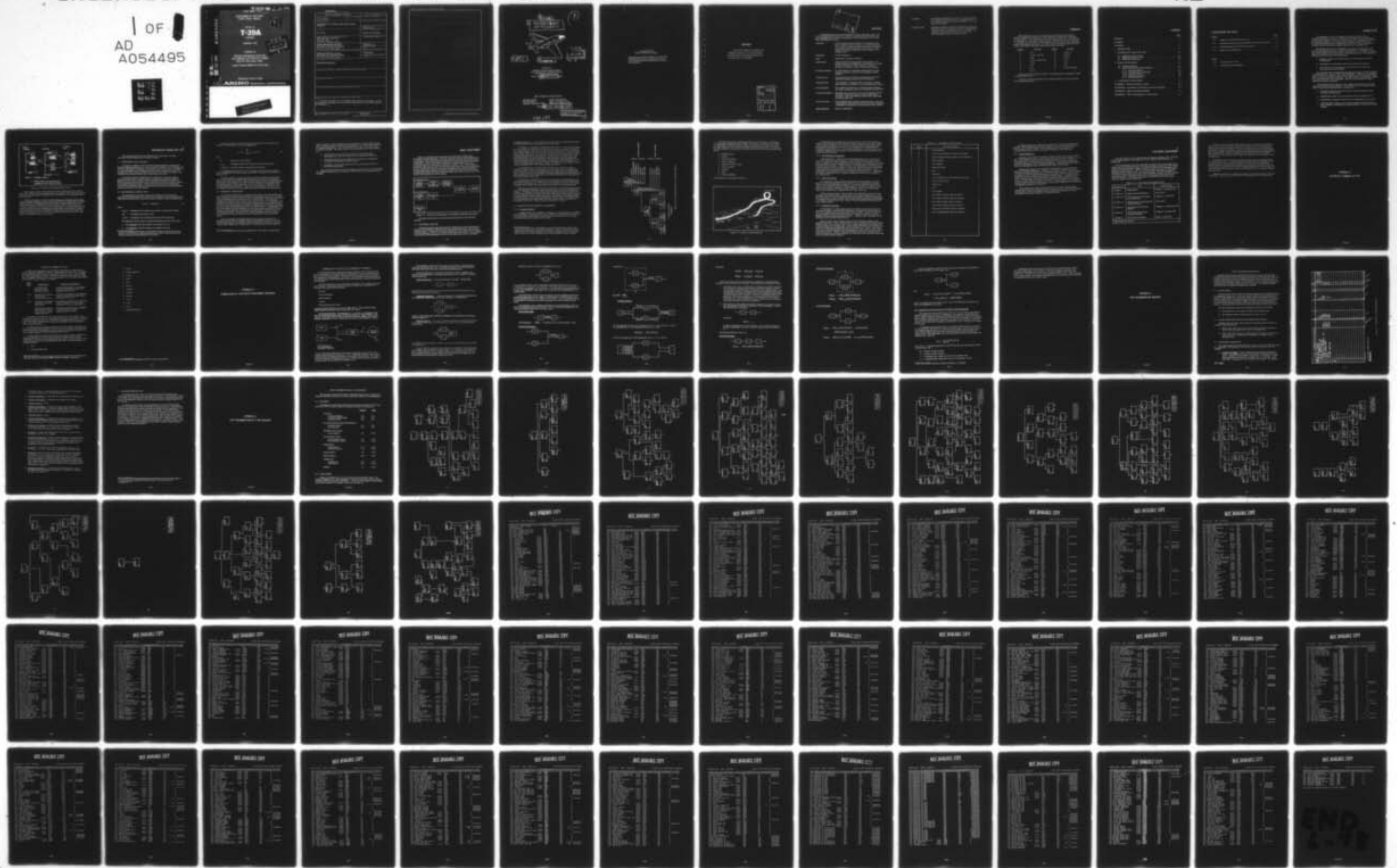
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Final Report A054494

DEVELOPMENT OF AIR FORCE
FLIGHT SAFETY MODELS

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Volume 14

T-39A

AIRCRAFT

September 1976

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MAY 31 1978

Prepared for

SERVICE ENGINEERING DIVISION
SAN ANTONIO AIR LOGISTICS CENTER
Kelly Air Force Base, Texas

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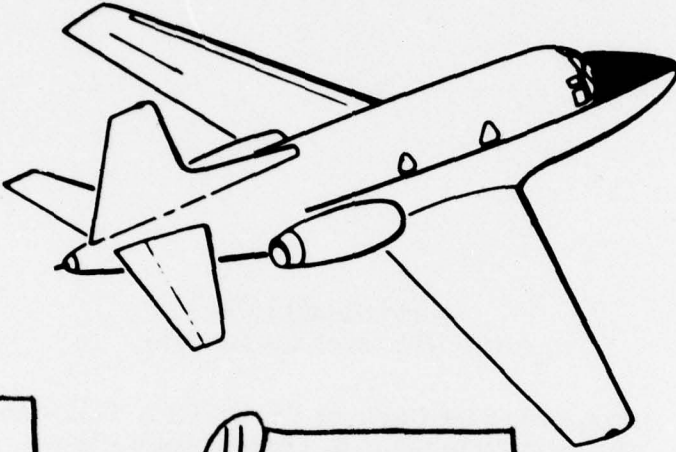
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Final Report

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FLIGHT SAFETY MODELS
Volume 14 T-39A AIRCRAFT



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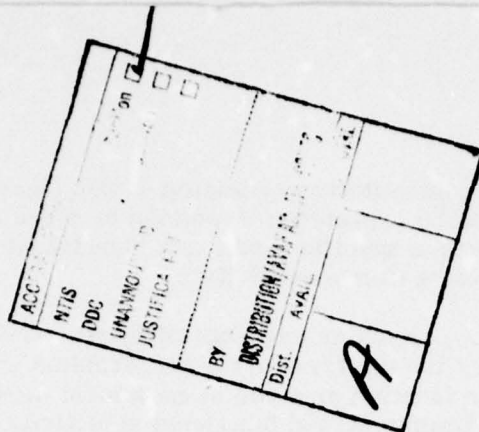
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ABSTRACT

A general description of the Flight Safety Prediction Technique, and the documentation associated with its specific application to the T-39A aircraft, are presented.

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GLOSSARY

This glossary presents general definitions of terms used in this report. The reader will find certain of these terms defined in somewhat different words in the text, depending on the context of the discussion; but the meaning will be consistent with the definitions given here.

- Criticality** - A numerical index of the significance of equipment failure history relative to aircraft safety. As an analysis parameter, it can be considered proportional to the likelihood that an item will fail and thereby cause an accident. It is the product of the failure probability and the sensitivity of an equipment item.
- Dependency** - See link dependency.
- FSPT** - Flight Safety Prediction Technique
- Flight Phases** - Discrete segments of the aircraft mission profile. For present purposes, the flight phases are defined as 1) startup and taxi, 2) takeoff, 3) climb, 4) cruise, 5) tactics, 6) cruise, 7) descend, 8) land, and 9) taxi and shutdown.
- Functional Analysis** - The determination of equipment relationships to aircraft functions performed, and the interrelationships of these functions.
- Functional Link** - The simplest form of functional relationship in which one function is dependent upon the next lower function.
- Functional Path** - The compilation of functional links, in sequence, through which a function is identified as being dependent upon another.
- Link Dependency** - The conditional probability of a dependent function failing, given that a particular function it is dependent upon has failed.
- Provisory Condition** - Operation of an aircraft in a mode or environment such that the safety-related importance of certain equipments is increased. Provisory conditions include icing, night flight, supersonic flight, etc.
- Provisory Factor** - The probability that a provisory condition exists. Also used to describe the coded notation used to indicate that a functional relationship is dependent on a particular provisory condition.
- Safety Sensitivity** - Same as "sensitivity".

Sensitivity

- A quantitative indication of the degree of safety degradation to be expected if a function or piece of equipment fails. The more specific terms are "functional sensitivity" or "equipment item sensitivity".

Sensitivity Path

- A particular sequence of functional dependencies (beginning at the top level in the hierarchical structure) through which a function or piece of equipment derives a sensitivity value. Equipment and functional sensitivity values are often derived through several such sensitivity paths.

FOREWORD

This document is part of a 16-volume report describing the application to specific aircraft types of ARINC Research Corporation's Flight Safety Prediction Technique (FSPT). The technique was developed under previous Air Force contracts (see Appendix A). The present effort, undertaken in 1972 under Contract F09603-72-A-1132-SA01, has led to further refinement of the FSPT through its broad application to many different types of aircraft. The flight safety models generated for these aircraft are presented in individual volumes of this report as follows:

<u>Volume</u>	<u>Aircraft</u>	<u>Volume</u>	<u>Aircraft</u>
2	T-38	10	B-52G, H
3	F-111A, FB-111A	11	C-130E
4	A-7D	12	KC-135
5	F-4D, E; and RF-4C	13	C-5A
6	C-141	14	T-39
7	A-37	15	F-15
8	O-2	16	UH-1N Helicopter
9	OV-10		

Volume 16 will document the results of a feasibility study of extending the FSPT to rotary-wing aircraft.

Volume 1, an overall summary of the contractual effort, will be issued at the end of the contract period.

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1 INTRODUCTION

The Flight Safety Prediction Technique developed by ARINC Research Corporation provides for assessment of the impact on flight safety of the failure of specific items of equipment within an aircraft. In the FSPT, mathematical modeling procedures are applied for processing aircraft-equipment failure data to yield a quantified index ranking safety-related problems on the basis of their likelihood of occurrence and the resulting degradation in the aircraft's capability to fly.

The ranking factor is called "criticality", which in its simplest form is the product of the failure probability and flight-safety sensitivity of an equipment. (A more detailed definition appears in Section 2 and Appendix B.) The failure probability inputs are from basic failure-data sources, AFM 66-1 and 65-110. The sensitivity estimates are derived by the following process:

- a. Systematic analysis of aircraft functions to determine those essential to flight safety
- b. Identification of the hardware required to perform these functions
- c. Evaluation of the safety significance of the hardware in performing these essential aircraft functions.

The criticality values resulting from this approach provide a relative ranking of all malfunctions with respect to their safety significance. Figure 1-1 is a simplified example of how three equipment items would be ranked on the combined basis of their failure probability and safety sensitivity. This figure illustrates an example in which item A has the highest failure probability, but due to the low sensitivity value is ranked below item B in criticality.

The methodology has the ability to rank malfunction problems currently and continuously by their accident potential. This ranking, based on criticality assessment, can provide the basic parameters necessary for:

- a. Identifying equipment items whose failure history and application pose a threat to aircraft safety
- b. Quantifying the degree of threat associated with each equipment item
- c. Evaluating and tracking the effectiveness of modifications to the aircraft
- d. Assessing safety benefits versus the cost of proposed aircraft modifications, changes in maintenance or flight operations, or alternative aircraft designs.

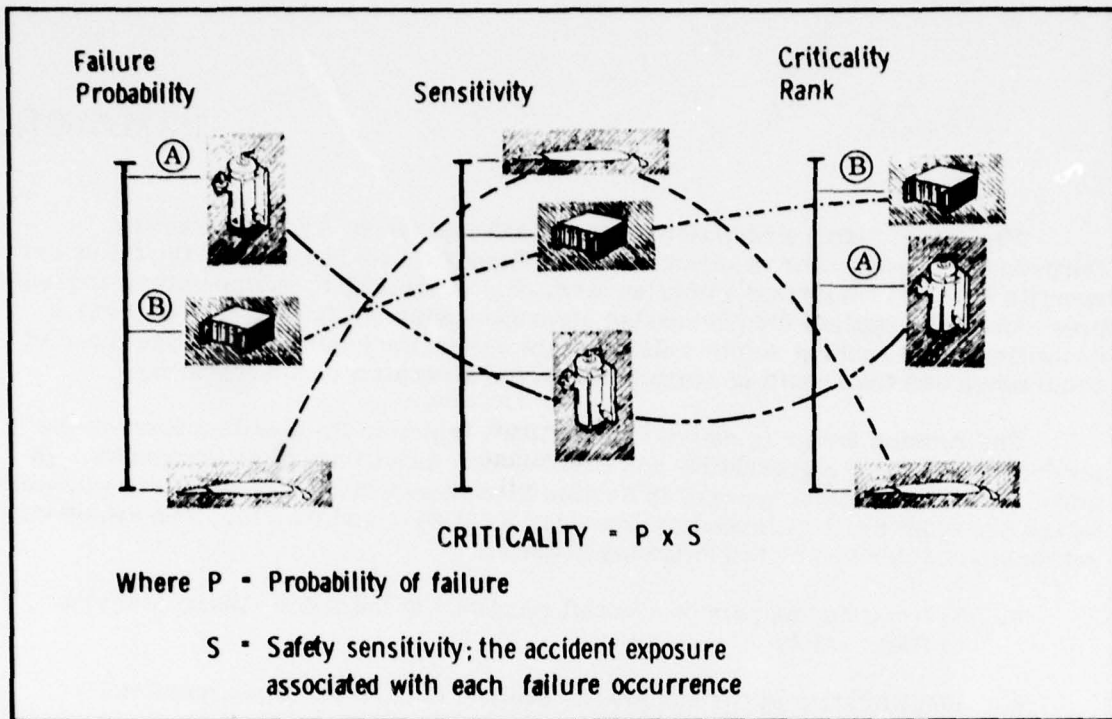


Figure 1-1. Example of Criticality Ranking Process

In this report, Section 4 and Appendix D pertain specifically to the T-39A aircraft. The remainder of the document provides support information that will make the T-39A data, and the method by which the data were obtained, more meaningful to the general reader.

Section 2 presents an overview of the development and utilization of the Flight Safety Prediction Technique; Section 3 discusses the steps associated with generating a safety model for calculating the safety criticality of various equipments of an aircraft; and Section 4 describes how the safety models for the T-39A aircraft were developed. Appendix A summarizes the contractual history of the development of the FSPT; Appendix B discusses mathematical considerations underlying the technique; Appendix C discusses FSPT documentation methods; and Appendix D presents functional relationship diagrams for a listing of keypunch cards that comprise the safety model documentation for the T-39A aircraft.

METHODOLOGY UNDERLYING FSPT

This section discusses the basic definitions and mathematical concepts associated with the Flight Safety Prediction Technique.

2.1 DEFINITION OF SAFE AIRCRAFT

To develop a relative measure of aircraft safety degradation resulting from specific equipment malfunctions, it is first necessary to define a "safe" aircraft. For purposes of the FSPT assessments, an aircraft is assumed to be in a safe condition if it is operating within its prescribed performance limits. Conversely, an aircraft operating (or about to operate) outside these limits is considered to be unsafe - in a condition where property damage and personal injury may result.

The safety prediction methodology does not attempt to assess the extent of possible personal injury or aircraft damage resulting from an unsafe condition. Neither does the concept consider ejection capability, parachutes, life rafts, etc., which do not make an aircraft safer per se but provide for the survivability of the aircrew when the aircraft is unsafe. Collision is also excluded from consideration because of the complexity of the interrelationships between pilot, aircraft equipment, ground surveillance, and traffic density.

2.2 MATHEMATICAL BASIS OF FSPT

The probability of an accident caused by the failure of an element can be expressed as the probability of the element failing multiplied by the conditional probability that the failure of the element will cause an accident. Stated in equation form:

$$P(A, j) = P(j)P(A|j) \quad (1)$$

where

$P(A, j)$ = Probability of an accident due to failure of just the j^{th} element*

$P(j)$ = Probability that element j fails

$P(A|j)$ = Probability of an accident given that the j^{th} element fails.

This equation reflects the basic relationships addressed in the FSPT where:

a. The criticality of the j^{th} element is an estimate of $P(A, j)$

b. The sensitivity of the j^{th} element is an estimate of $P(A|j)$

*In this and subsequent discussions, unless otherwise stated, expressions such as "failure of the j^{th} element" should be interpreted to mean: failure of only the j^{th} element, assuming all other elements are not failed.

Because an element's effect on safety may depend on the mission phase (see Section 3.2.1), the above model can be expanded to:

$$P(A, j) = \sum_{k=1}^N P_{j,k} P(A|j,k) \quad (2)$$

where

N = Number of mission phases

$P_{j,k}$ = Probability that the j^{th} element is failed in the k^{th} phase

$P(A|j,k)$ = The j^{th} element's sensitivity in the k^{th} phase.

To identify the importance of discrete elements to aircraft safety, a flight profile consisting of nine distinct phases was defined. The phases are discussed in Section 3.2.1.

To utilize equation 2, it was necessary to develop a method for obtaining the values of $P(A|j,k)$, the probability that a malfunction in element j during mission phase k will result in an accident. This method in turn requires the estimation of two parameters: the probability of accident if a major function is not available during each mission phase, and the dependence of the major function on subfunctions and elements during each such phase*. Each function and equipment item thus derives its sensitivity value from its relationship to the major function(s) dependent upon it.

2.3 SENSITIVITY ASSIGNMENTS

A great deal of information is available on the causes of aircraft accidents, but little exists from which to make the sensitivity assignments $[P(A|j)]$. These assignments are therefore largely subjective, based on the analyst's knowledge of the system and any information he may have on previous accident history. The sensitivity assignments are reviewed (and revised as necessary) by an Air Force/contractor team working on a particular model to ensure that consistent criteria have been followed. The team review and negotiation of sensitivity assignments is the mechanism by which the value becomes sufficiently objective for use with the model. This negotiation considers all of those top level functions as a group and reassigns sensitivity values as necessary to assure that the most objective proportionality is attained for the particular aircraft model. The same major-function sensitivity values are used for major functions on all aircraft models where configuration and mission profiles permit.

The development of criticality rankings for the various elements (j 's) is dependent upon the ability to quantify the failure probability $[P(j)]$ and the element sensitivity $[P(A|j)]$ for each element. Since the intent of the concept is to provide a relative safety ranking of all malfunctions, it is not necessary to develop absolute

*For a more detailed discussion of the mathematics of the FSPT, see Appendix B.

values for $P(A|j)$. If the sensitivity values developed are correct relative to each other, a proper criticality ranking will be established. It is intended that criticality be an index proportional to $P(A, j)$ and therefore provide the same relative rank ordering of elements. The major reasons for proportionality, rather than equality, are:

- a. The FSPT does not account for the effect of extraordinary pilot intervention to prevent an accident in case of equipment malfunction.
- b. Criticality quantification was limited in its treatment of simultaneous occurrence of independent, primary failures.
- c. Operational and malfunction data yield only a proportional estimate of the required information.

While strict proportionality cannot be mathematically proven, it is believed that the criticality rankings provide reasonable relative measures of equipment problem potential.

Figure 3-1 summarizes the approach to the assessment of flight-safety criticality of aircraft equipment. The first contractor activity is the identification of all functions the aircraft is expected to perform and the determination of their inter-relationships. Next, each functional relationship is documented; and then sensitivity assignments are made at the major functional levels (below these levels, link dependency values are estimated; see discussion, Section 3.2.2). This process is carried out until each work unit code associated with a major function has been identified with respect to the function performed and dependencies have been estimated. Computer processing calculates the safety sensitivity for each work unit coded item, combines these values with the operation and failure data input by the Air Force, and produces the equipment criticality ranking.

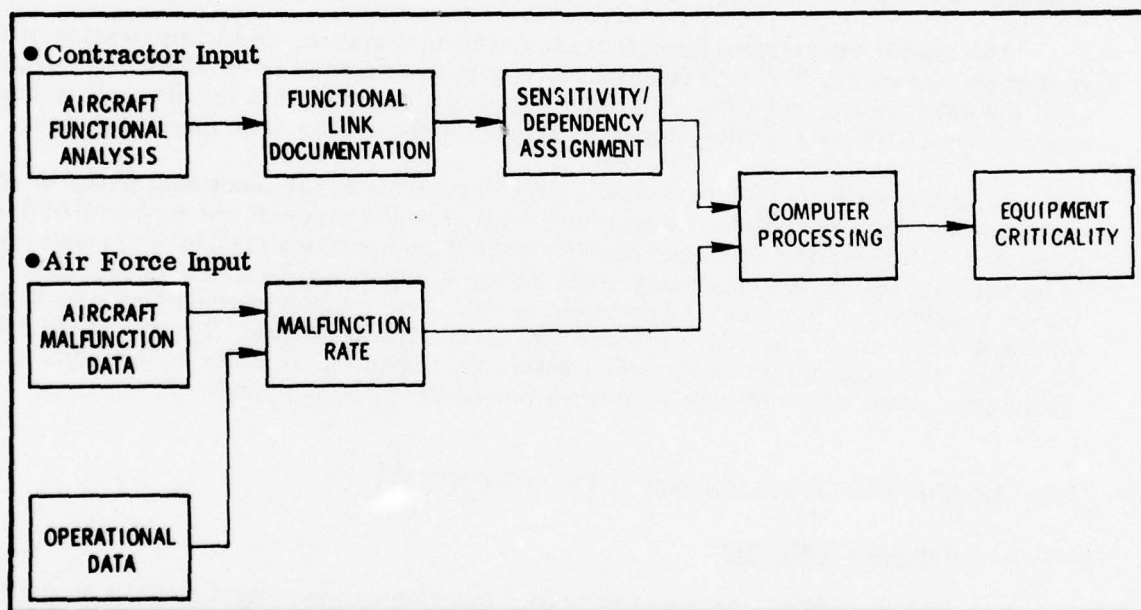


Figure 3-1. Activities and Data Inputs to Flight Safety Criticality Assessment

The steps in this process are discussed in greater detail in the following sections.

3.1 FUNCTIONAL ANALYSIS

Functional analysis entails the systematic identification of the relationships of hardware to the functions performed by the aircraft and documented in the aircraft technical orders. Tabulated for each aircraft function are the equipments necessary for its performance as well as all outputs required for other systems. The complexity of the functional interdependencies of an aircraft requires the use of a systematic

accounting procedure, as discussed below, to assure that all relationships have been identified and that no functional paths have been overlooked.

Certain top-level functions (comprised of both "primary" and "major" functions) have been defined as applicable to all aircraft types, and serve as the starting point for a safety analysis. Figure 3-2 lists these top level functions with the primary function of Flight Control expanded to show its typical major functions. Below the major function level, differences in aircraft types result in function identification and structuring specifically suited to each aircraft. In Figure 3-2, for instance, the major function Roll Control is subdivided into Left Roll and Right Roll, and further into aileron and spoiler actuation subfunctions. This structure is that applicable to an F-4 aircraft, in which ailerons have an extremely limited upward travel and lift is primarily lost through spoiler operation. Finally, each item in the aircraft WUC ("-06") manual is identified with respect to the function it performs. *

Every function and every WUC included in the model receives an "alpha designator" unique to that aircraft model. Due to the large number of alpha designators required in a model, an indenturing system is utilized to prevent duplication. However, the location in the hierarchical structure and the number of characters in the alpha designators are often independent, since such correlation is not necessary for subsequent computer processing.

The functional relationships from the system diagram, and identification of the equipment necessary for each function, are next documented in an 80-column punch-card format (see Appendix C). The total functional diagram for the aircraft is then a compilation of the system diagrams, with one punchcard for each functional link.

With the aircraft functions completely documented, the functional paths by which a piece of equipment contributes to the operation of the aircraft can be identified by computer. Performing the path-identification/documentation task by computer proves to be not only useful but necessary - the human analyst could neither keep track of nor assign sensitivity values to all functional paths. The machine processing capability allows the analyst to consider only one functional link at a time. The ability to follow all of the functional interrelationships within the aircraft, which is necessary for meaningful assessment of safety, is then provided by the computer.

3.2 MAJOR-FUNCTION SENSITIVITY ASSIGNMENT

3.2.1 Assignment Method

As stated earlier, the sensitivity of a function or equipment item is an estimate of the probability that its failure will cause an accident. From functional analysis of the aircraft under consideration, major functions are identified and are assigned sensitivity values for each phase of the mission.

*Certain WUC items in the "-06" manual may not be included in the safety model, these items being either 1) eliminated by TCTOs; 2) purely structural items in the 11000 series; 3) necessary only for survivability or ejection; 4) of lower indenture than the LRU level, where computer data screening eliminates failure reports.

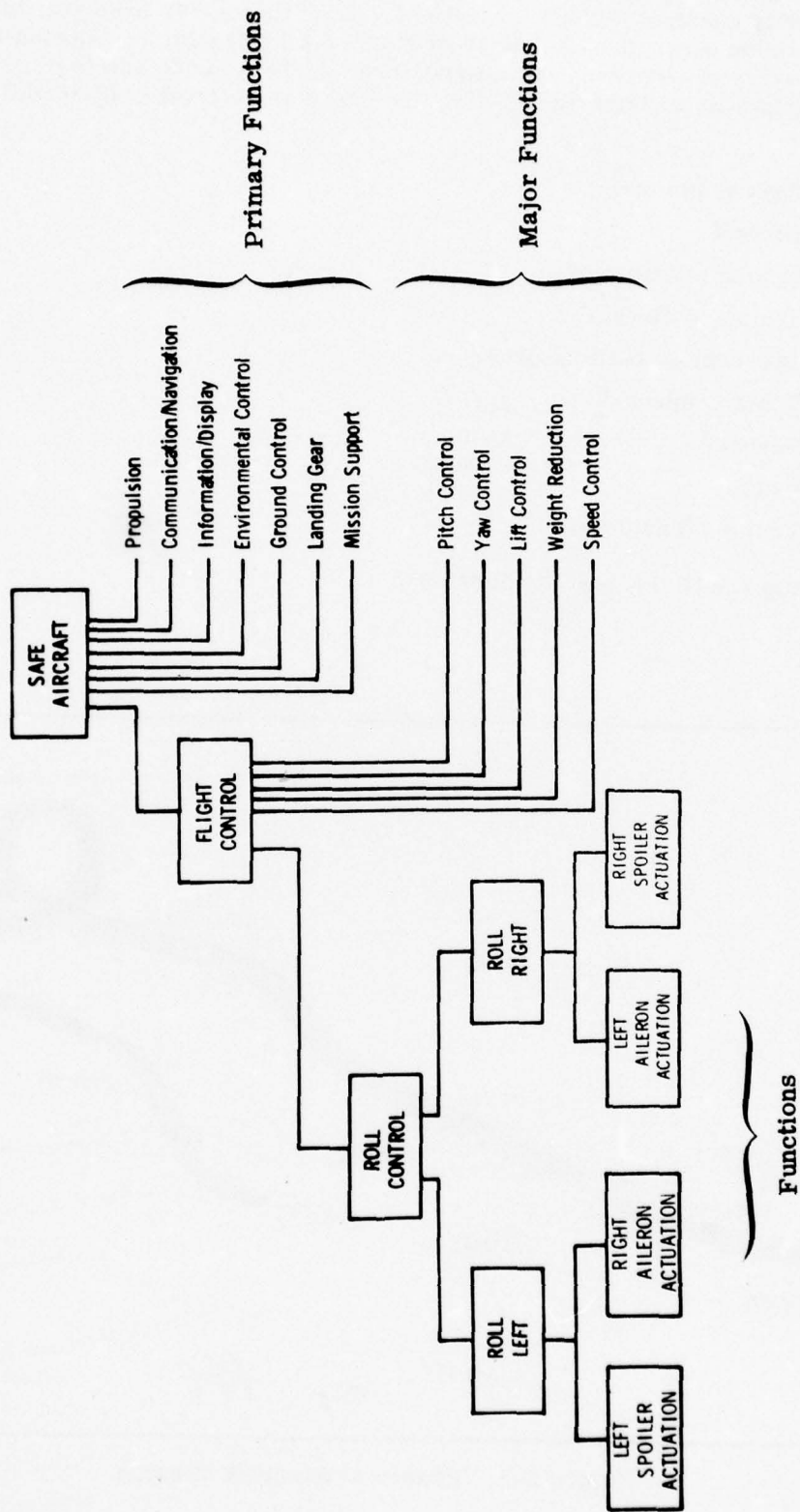


Figure 3-2. Hierarchical Structure of Aircraft Functions

The relative importance of primary functions, major functions, and functions is not necessarily constant throughout a flight. The failure, for example, of one engine of a multi-engine aircraft is far more critical on takeoff than it is during the rest of the flight, and is of relatively little importance during startup and taxi. To accommodate this variability of importance, the mission of an aircraft is divided into nine flight phases:

1. Startup and taxi
2. Takeoff
3. Ascend (climb-out)
4. Cruise, outbound
5. Intercept or tactical phase
6. Cruise, inbound
7. Descend
8. Land
9. Taxi and shutdown

These phases are illustrated in Figure 3-3.

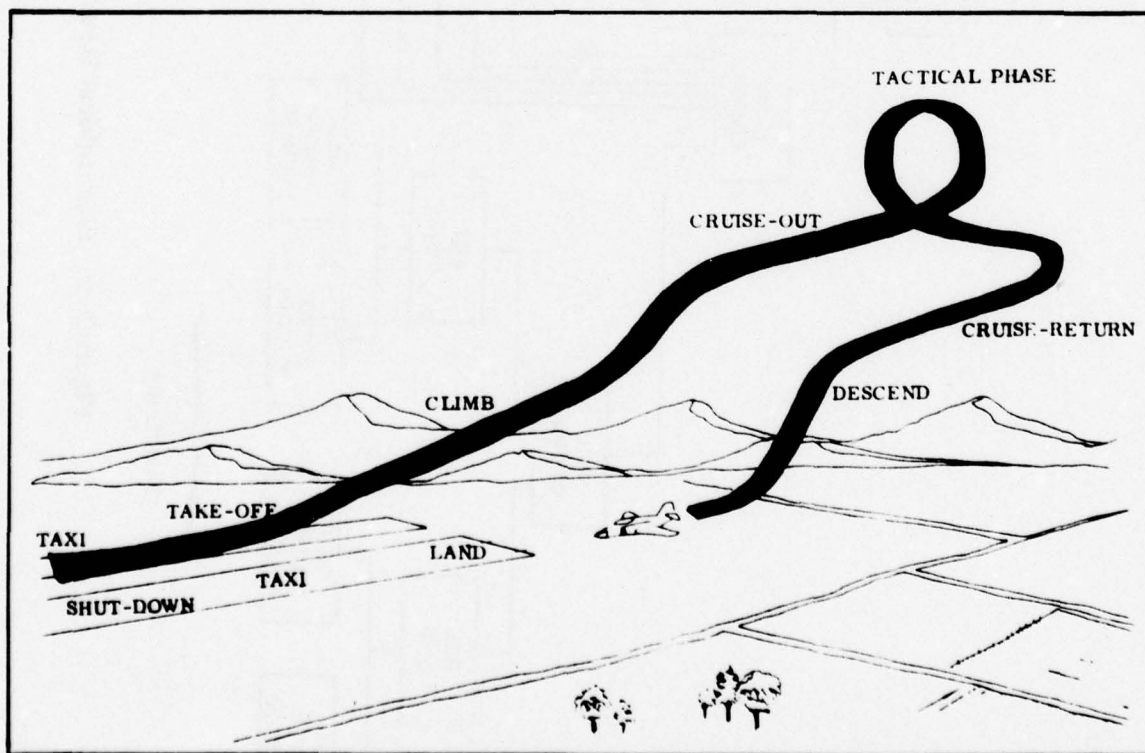


Figure 3-3. Phases of Aircraft Mission

A sensitivity value is assigned for each of the phases, and represents the best estimate of the likelihood that the aircraft will enter a hazardous mode if the function is not present in that phase. The numerical values assigned are proportional rather than absolute, and range from 0.0 to 1.0. The keypunch card format limits this assignment to increments of 0.1. Increments smaller than 0.1, when required, were assigned by defining a quasi-function for insertion between the major function and its dependent primary function.

3.2.2 Link Dependency Assignment

"Link dependency" is defined as the probability that the loss of a function will result in the loss of a dependent function. (For a more detailed discussion of this term, see Appendix B.) The assignment of link dependency values requires knowledge of the operation of specific aircraft because it is concerned only with functional levels below the "major" category. At this lower level, no evaluation is made of the impact on flight safety of the loss of functions. Instead, the effect of the loss of one function on the performance of another function becomes the evaluation criterion. Like sensitivities, link dependency values are assigned in increments of 0.1. Additionally, the method of attenuation used in assigning sensitivity values can also be applied to link dependencies.

3.2.3 Provisory Factors

The sensitivity of major functions with respect to aircraft safety, and at the lower levels the link dependency between functions, can be dependent on external influences and aircraft operating conditions. To accommodate these external influences, a set of provisory factors has been identified. An example would be a windshield anti-ice system, which has a safety sensitivity close to 1.0 during landing under icing conditions but a negligible effect on a dry, warm day.

Under such circumstances, the procedure is to assign the "worst case" value (assuming the condition exists). During model exercise the likelihood that the condition exists can be "read-in", thereby allowing the sensitivity value to be assigned by the computer based on the likelihood of the condition and the probability that the higher level function will therefore be lost. Table 3-1 lists the standard provisory factors used in FSPT models.

3.2.4 Computer Processing

Documentation of a flight safety analysis by ARINC Research thus consists of functional diagrams, coded functional tabulations, a functional data processing card deck, and a machine-prepared printout of the card deck data. Under this contract, the documentation is then sent to San Antonio Air Logistics Center for review by MMER personnel and representatives of the Air Logistics Center responsible for the particular aircraft (if other than SA/ALC).

SA/ALC processes the functional data card deck utilizing a number of computerized operations. First, a functional deck edit is accomplished to identify certain format or logic errors that may exist. Next, a path identification/documentation run is made that traces all possible paths associated with each function and calculates the numerical sensitivities by flight phase down to the WUC level. Then, a path combination run is made taking into account the dependence of more than one major function on a particular WUC. Finally, failure information from the 66-1 data system and numerical factors for provisory conditions are input and a WUC criticality list by rank order is generated by the computer.

TABLE 3-1. PROVISORY FACTOR CODES

Code	Provisory Condition
A	Icing conditions
B	Adverse speed/altitude operations (Helicopter)
C	Runway stopping distance/confined area (Helicopter)
D	Night operation
E	IFR conditions
F	Supersonic flight
G	Rain
H	Solo flight
I	Loss of function for which indication is provided
K	Normal system failed
T	Flame-out
X	Fire
Y	Cold weather
2	One of three available units is required
3	Two of three available units are required
4	One of four available units is required
5	Two of four available units are required
6	Three of four available units are required
8	Four of eight available units are required

An additional product generated by the computer is a two-part criticality trend analysis. Part I contains the criticality rankings and linear regression analysis by WUC for the previous 12 months. Part II contains plots of the criticalities and regression lines for the 25 WUCs top-ranked according to safety criticality.

3.2.5 Model Maintenance

Each time an aircraft type for which a safety model has been developed undergoes a modification, the effects of the changes on the model must be evaluated. Technical order and WUC revisions must be incorporated into the model. Removal of existing hardware, the installation of new hardware, or design improvements may change link dependencies and sensitivity assignments. The update procedure should follow the same general steps as outlined for the initial analysis effort.

Existing block diagrams and a printout of the functional card deck form the baseline for change identification. Functional relationships should be reviewed to determine the impact of changes on the documented safety analysis. Diagrams should be revised to reflect functional differences, WUC changes should be noted, and all differences listed on a flight-safety functional tabulation sheet. The functional deck printout can be used for manual indication of what the changes are and where they occur. New data cards are prepared and the functional deck updated by the removal of obsolete cards and the insertion of new cards. From this point on, the computer is again utilized to edit the functional deck, perform path identification/documentation, and calculate sensitivities for each WUC.

Block diagrams and other affected portions of the specific aircraft safety analysis report should be updated and revised pages issued that reflect these changes. Maintaining an accurate and updated model is important to obtaining an accurate assessment of the safety significance of hardware failures.

T-39A MODEL DEVELOPMENT

The FSPT model for the T-39A aircraft was begun in September 1975. The total aircraft documentation was submitted for "GO-95" computer edit at SA/ALC in August 1976.

The aircraft flight manual and maintenance technical orders provided the information on aircraft system operation. The model developed represents T-39A aircraft configured to the latest time compliance technical orders (TCTOs) documented in the manuals supplied by SA/ALC. Table 4-1 lists the manuals and their revision status applicable to the developed model. As noted in the table, two technical manuals (1T-39A-2-3 and 2-4) were not received in their entirety. However, SA/ALC supplied reproduction copies of applicable sections from the manuals which were used for model development. The reproduction copies are assumed to be the most recent revisions, which occurred during 1975.

TABLE 4-1. T-39A SYSTEM DOCUMENTATION

Nomenclature	Title	Revision/Date
1T-39A-1	Flight Manual	Change 1, 31 January 1975
1T-39A-2-3	Engines and Related System	Circa 1975*
1T-39A-2-3-1	Power Plant Ground Operation and Conditioning	Change 18, 2 June 1975
1T-39A-2-4	Flight Control and Hydraulically Operated Systems	Circa 1975*
1T-39A-2-5	Instruments and Electrical Systems	Change 17, 19 February 1974
1T-39A-2-6	Radio Communication and Navigation Systems	Change 21, 30 April 1975
1T-39A-06	Work Unit Code Manual	Basic, 1 April 1975
*See discussion, Section 4.		

Because of the vulnerability of the functional logic/sensitivity documentation to such errors as omission of links, duplication of cards, and incorrect keypunching, quality reviews were conducted at various critical points in the model development. In addition to keypunch verification, each card was checked against the functional link shown on the original rough draft and the final functional diagram and the diagrammed link was checked off. Missing or duplicated functional links were thus identified. Work unit codes used in the model were checked off against the WUC manual to assure completeness.

The quality reviews were first conducted prior to computer verification of the aircraft deck by SA/ALC. Following computer verification, a second quality review was performed by representatives of Warner Robins ALC and ARINC Research. Finally, the first criticality printout obtained from application of actual aircraft data was reviewed to identify any terms whose sensitivity appeared to be unreasonable. In such cases the paths were traced manually and changes made if an erroneous relationship was found.

Appendix C presents the methods and standard used in documenting an FSPT aircraft model. Appendix D presents the FSPT documentation for the T-39A aircraft.

APPENDIX A
HISTORICAL SUMMARY OF FSPT

HISTORICAL SUMMARY OF FSPT

In 1965, the desirability and practicability of quantifying the significance of specific equipment malfunctions relative to flight safety was explored in a feasibility study conducted by ARINC Research Corporation for the Air Force. The feasibility of a safety-quantification approach, which has subsequently become known as Flight Safety Prediction Technique (FSPT), was demonstrated; and the method was developed and refined in a series of studies, as follows:

<u>Study Phase</u>	<u>Subject/Date</u>	<u>Sponsor*/Publication No.</u>
I	Feasibility Study, September 1965 to June 1967 (Phase I)	Sacramento Air Materiel Area (SMNE), Contract AF09(603)62335, SM-37-2; publication 705-01-1-777
II-A	Technique Development, October 1967 to July 1968 (Phase II-A)	San Antonio Air Materiel Area (SANEW), Contract AF09(603)-67-A-0267-SA01; publication 734-01-1-895
II-B	Technique Development, July 1968 to July 1969 (Phase II-B)	San Antonio Air Materiel Area (SANEW), Contract F09(603)-68-A-0317-SA01; publication 754-01-1-985 (Revision 1)
	FSPT System Documentation for the F-4C and T-37 Aircraft, October 1970 to June 1971	San Antonio Air Materiel Area (MMER) Contract F41608-71-C-0576; publication 697-01-1-1118

In the Phase II-B study, the FSPT was applied to the F-106 aircraft. Concurrent with Phase II-B, the U.S. Naval Safety Center contracted ARINC Research to extend the methodology to produce a flight safety criticality model for the F-4J aircraft. The results of this effort are documented in ARINC Research Publication 753-01-3-982 (Revision 1).

In 1970, ARINC Research was contracted to develop suitable input data to permit the application of the technique to the T-37 and F-4C aircraft. These data were derived in the form of mathematical model functional documentation as input to the basic computer program developed and applied to the F-106.

In 1972, ARINC Research Corporation was awarded a contract, with the subsequent modifications in 1973 and 1974, to apply the Flight Safety Prediction Technique to 15 aircraft, working jointly with cognizant Air Logistics Centers. Aircraft to which the FSPT has been applied under this latter contract (F09603-72-A-1132-SA01) include:

- a. T-38
- b. F-111A and FB-111A

*The office symbols of Service Engineering at the Sacramento and San Antonio Air Materiel Areas are now SM/ALC/MME and SA/ALC/MME, respectively.

- c. A-7D
- d. F-4D, E; RF-4C
- e. C-141
- f. A-37
- g. O-2
- h. OV-10
- i. B-52G, H
- j. C-130E
- k. KC-135
- l. C-5A
- m. T-39
- n. F-15
- o. UH-1N Helicopter*

*Feasibility study of adaptation of FSPT to rotary-wing aircraft.

APPENDIX B
FORMULATION OF CRITICALITY-ASSESSMENT TECHNIQUE

FORMULATION OF CRITICALITY-ASSESSMENT TECHNIQUE

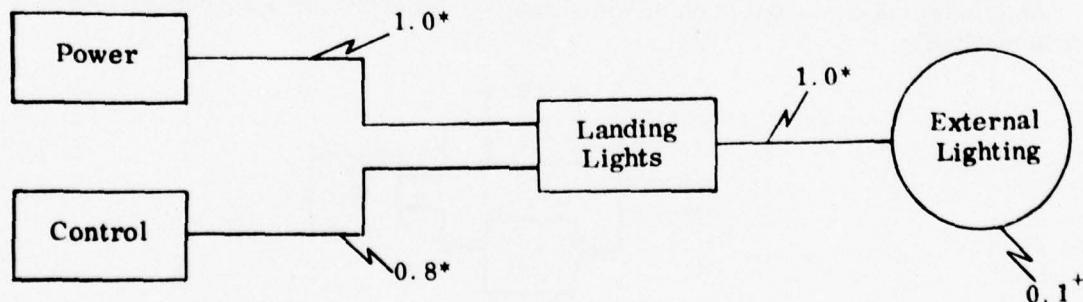
To implement the basic safety model defined in Section 2.2, it is necessary to develop a submodel for the probability that a malfunction in element j during mission phase k will result in an accident. This submodel in turn requires that we estimate two parameters: the probability of accident if a major function is not available during each mission phase, and the dependence of the major function on element j during each mission phase.

The first parameter is termed "functional sensitivity" and is estimated for each major function. The functional analysis performed in this task established for an aircraft the following hierarchal scheme:

- Aircraft
- Primary functions
- Major functions
- Function
- Elements (Work Unit Codes)

A primary function would be one such as Flight Control. Major functions under Flight Control would include Pitch Control and Yaw Control.

The second parameter, "link dependency," is a vehicle for showing the influence of each functional-path element on the performance of a major function. For example, if the major function being considered is External Lighting, the following diagram illustrates the nature of functional sensitivity and link dependency values.



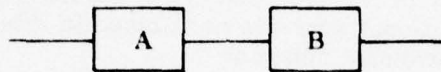
- * Link dependencies
- + Functional sensitivity

The 0.8 value means that failure of the Control function will result in loss of the Landing Light function 80% of the time. The 0.1 functional sensitivity value denotes that loss of external lighting will result in an accident 10% of the time. The values must be interpreted in a proportional sense, in that the actual accident probability is dependent upon external factors (see Section 3.2.3).

The remainder of this appendix discusses the procedures and model used to obtain element sensitivities; e.g., in the above example, the accident probability given that a Work Unit Code in the Control function malfunctions.

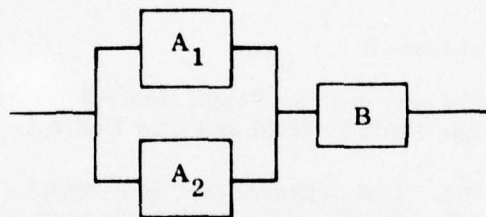
Three principal types of functional relationship--series, redundant, and parallel--were identified as representing the major forms to consider in modeling element sensitivity.

Series Relationship - A function having only one input. Schematically,



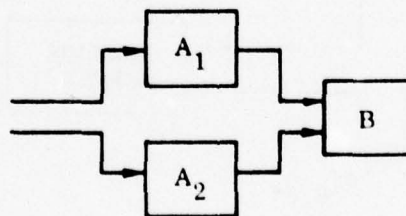
which indicates that outside of its own elements, the success of function B is only affected by the success of function A.

Functional Redundancy - A function having one or more backup functions that can provide the required inputs to successor functions. Schematically,



where A_1 and A_2 represent a functional redundancy in that either may provide the necessary input to B.

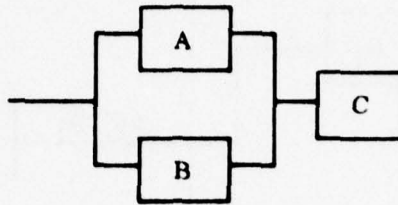
Parallel Functions - Two or more functions independent of each other in terms of functional success, but each of which may be required for a successor function. Schematically,



B will generally require both A_1 and A_2 ; but A_1 does not depend on A_2 , nor does A_2 depend on A_1 .

In some cases the distinction between functional redundancy and parallel paths is very slight, and may depend on mission phase. For example the four engines of a plane can be considered to be a redundant configuration providing inputs to the primary propulsion function during cruising, but would generally be considered to be parallel functions during takeoffs requiring full power.

In general, given a schematic relationship of the form,

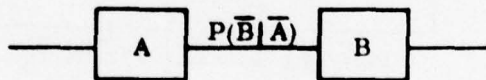


we can say that A and B are in a functionally redundant configuration if the success probability of C is the same if 1) A and B are successful, 2) A only is successful, or 3) B only is successful. If, for example, C is more likely to be successful if both A and B are successful, rather than A or B alone, then the relationship is one of parallel paths.

It is noted that the model will also account for element redundancy and parallel elements through inputs such as $P(\bar{A}|i_a)$, representing the probability that the Ath function fails given that the i_a^{th} element in A has failed. If i_a is a parallel element, the probability would depend on mission requirements and other parallel-element states.

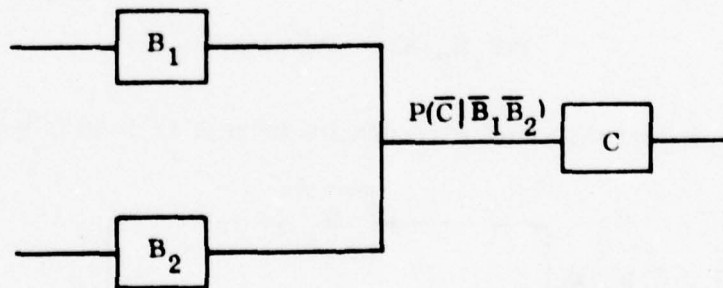
Link dependency is the conditional probability of a functional failure, given the failure of immediate predecessor functions. The link dependencies applicable to the three basic designs defined above are shown below.

Series Relationship

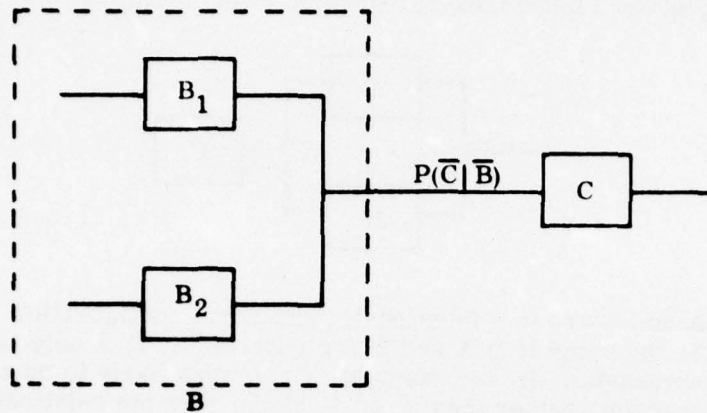


Link dependency = $P(\bar{B}|\bar{A})$ = probability that B fails given that A fails.

Functional Redundancy

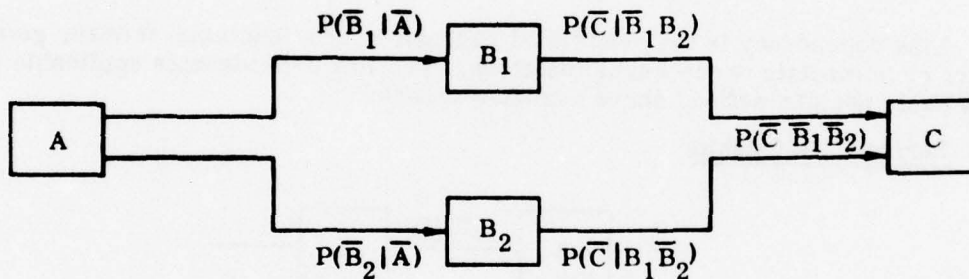


equivalent to



where $\bar{B} = \bar{B}_1 \bar{B}_2$

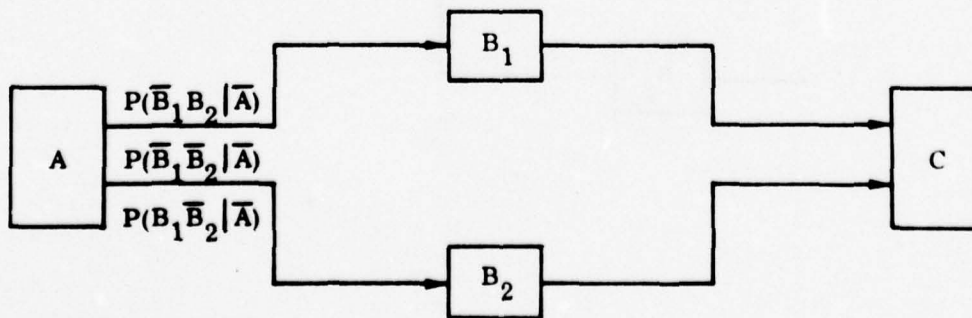
Parallel Functions



We shall generally assume that the dependencies of B_1 with respect to A , and of B_2 with respect to A , are independent of each other, so that

$$P(\bar{B}_1 \bar{B}_2 | \bar{A}) = P(\bar{B}_1 | \bar{A}) P(\bar{B}_2 | \bar{A})$$

We then can consider three link dependencies from A to B as follows:



noting that

$$P(\bar{B}_1 | \bar{A}) = P(\bar{B}_1 B_2 | \bar{A}) + P(\bar{B}_1 \bar{B}_2 | \bar{A})$$

$$P(\bar{B}_2 | \bar{A}) = P(B_1 \bar{B}_2 | \bar{A}) + P(\bar{B}_1 \bar{B}_2 | \bar{A})$$

Models are shown below for determining the sensitivity of elements within a function for each of the three basic designs. The following basic assumptions apply:

- a. Except for cases where an element has a redundant or parallel counterpart or is located in a function with a redundant or parallel function, only the element under consideration shall be assumed to have failed initially. Thus the expression $P(A | i_a)$, representing the accident probability given failure of the i th Work Unit Code element, is based on the assumption that no other element has failed unless element i is in some redundant or parallel configuration. For cases in which there are redundant or parallel counterparts, failures of such counterpart elements or functions are considered in accordance with their occurrence probabilities.
- b. The success of all immediate predecessors ensures the success of a function, provided that the function experiences no element failures. Thus for the series function relationship



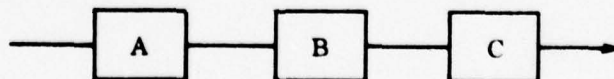
we assume

$$P(\bar{B} | A) = 0,$$

provided B experiences no element failures. If an element in function A is under consideration, the latter provision is always true by assumption "a."

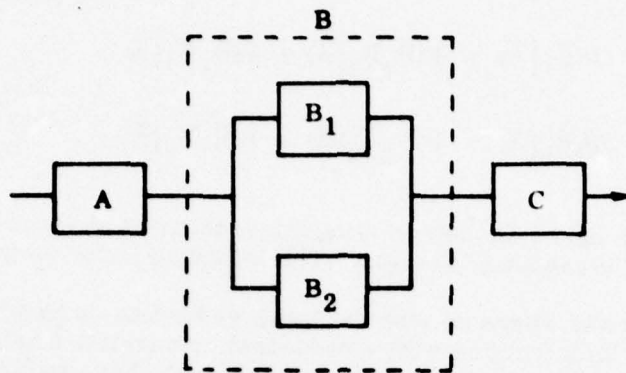
The element sensitivity models are:

Series Relationship



$$P(A | i_a) = P(\bar{A} | i_a) P(\bar{B} | \bar{A}) P(\bar{C} | \bar{B}) P(A | \bar{C})$$

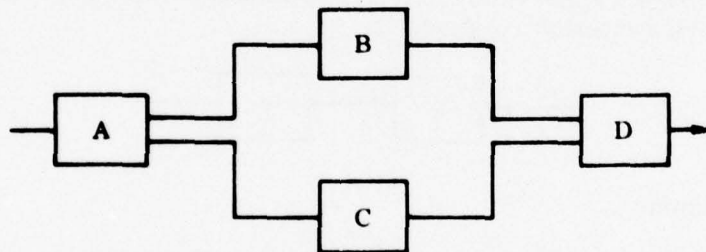
Functional Redundancy



$$P(\mathcal{A}|i_a) = P(\bar{\mathcal{A}}|i_a)P(\bar{\mathcal{B}}|\bar{\mathcal{A}})P(\bar{\mathcal{C}}|\bar{\mathcal{B}})P(\mathcal{A}|\bar{\mathcal{C}})$$

$$P(\mathcal{A}|i_{b1}) = P(\bar{\mathcal{B}}_1|i_{b1})P(\bar{\mathcal{B}}_2)P(\bar{\mathcal{C}}|\bar{\mathcal{B}})P(\mathcal{A}|\bar{\mathcal{C}})$$

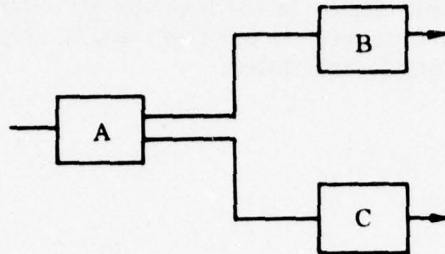
Parallel Functions



$$P(\mathcal{A}|i_a) = P(\bar{\mathcal{A}}|i_a) \{ P(\bar{\mathcal{B}}\bar{\mathcal{C}}|\bar{\mathcal{A}})P(\bar{\mathcal{D}}|\bar{\mathcal{B}}\bar{\mathcal{C}}) + P(\bar{\mathcal{B}}\bar{\mathcal{C}}|\bar{\mathcal{A}})P(\bar{\mathcal{D}}|\bar{\mathcal{B}}\bar{\mathcal{C}}) \\ + P(\bar{\mathcal{B}}\bar{\mathcal{C}}|\bar{\mathcal{A}})P(\bar{\mathcal{D}}|\bar{\mathcal{B}}\bar{\mathcal{C}}) \} P(\mathcal{A}|\bar{\mathcal{D}})$$

$$P(\mathcal{A}|i_b) = P(\bar{\mathcal{B}}|i_b) \{ P(\bar{\mathcal{C}}|i_b)P(\bar{\mathcal{D}}|\bar{\mathcal{B}}\bar{\mathcal{C}}) + P(\mathcal{C}|i_b)P(\bar{\mathcal{D}}|\bar{\mathcal{B}}\bar{\mathcal{C}}) \} P(\mathcal{A}|\bar{\mathcal{D}})$$

A case not explicitly included in the above three basic functional relationships is one for which a function is in two paths, e. g. ,



then

$$P(\mathcal{A}|i_a) = P(\bar{C}|i_a)P(B|i_a)P(\mathcal{A}|\bar{C}B) + P(C|i_a)P(\bar{B}|i_a)P(\mathcal{A}|C\bar{B}) \\ + P(\bar{C}|i_a)P(\bar{B}|i_a)\{1 - P(\bar{\mathcal{A}}|\bar{C})P(\bar{\mathcal{A}}|\bar{B})\}$$

where it is assumed that the effects of loss of the major functions in accident occurrence are independent of each other.

Use of Numerical Provisory Factors for Partially Redundant Systems

The numerical provisory factors (see Table 3-1) are used where more than two identical functions are involved in a redundancy. For example, aircraft with more than two engines often have identical and independent systems for hydraulic pressurization, and for electrical power generation, one driven by each engine. If the aircraft can be operated safely with one or more of such systems in a failed state, one of the numeric codes is utilized in assigning link dependency values. Consider, for example, the following:

If N identical and independent units* are available and at least M are required for safe operation, where $0 < M < N$, then the provisory factor of a given unit, say U_j , is the probability that the failure of U_j will cause the aircraft to enter an unsafe state. This is the probability that exactly $M-1$ of the remaining $N-1$ units will be in an unfailed state. This probability can be calculated by the formula for the binomial distribution, and is given by

$$P(U_j) = \binom{N-1}{M-1} p^{(M-1)} q^{(N-M)}$$

where $P(U_j)$ = probability that failure of the j^{th} unit will cause the aircraft to enter an unsafe state, and

M = Number of units required

N = Number of units available

p = Probability that a single unit will be in an unfailed state

q = Probability that a single unit will be in a failed state or $(1-p)$

*Units may be either elements, element assemblies, or functions.

Assignment of link dependencies to N identical and independent units of which only M are required proceeds as follows. The value assigned to each unit is the dependency of the higher level function on receiving an output from M of the units (usually 1.0). The provisory factor is the appropriate numeric code. In the evaluation of the path sensitivity, the computer is programmed to select the binomial formula that corresponds to the provisory factor listed.

APPENDIX C
FSPT DOCUMENTATION METHODS

FSPT DOCUMENTATION METHODS

Because of the extreme complexity of aircraft, it is necessary to develop a computerized method to identify and document all possible paths associated with each function as well as to determine the safety sensitivity associated with each path. A computer routine has been devised that takes the data from the functional card deck and traces and documents all paths. For each WUC, it also computes the flight-phase sensitivities for each path in which the WUC is present. The resulting computer printout provides a combined functional path sensitivity.

C.1 ALPHA CODING

As each system of the aircraft is functionally diagrammed, the functional blocks are assigned an "alpha code". This code aids the analyst in the bookkeeping tasks of functional diagramming and provides the computer with an identification of the elements to be processed. For standardization among aircraft, nine top-level functions have been defined and each has been assigned an initial or first-alpha designator. Each block in the functional diagram carries the same initial alpha as the top level function. Subsequent letters added to the initial alpha uniquely identify each block.

The only restrictions placed on the assignment of alpha codes are that:

- a. All characters in a code must be a letter of the alphabet, and
- b. The maximum number of characters in one code is seven.

C.2 ALPHA CODING AND COMPUTER PROGRAM COMPATIBILITY

Additional rules for alpha coding required to obtain the desired results from computer processing include:

- a. When a WUC item operates in the same mode to perform more than one function, the same alpha code is used in each application.
- b. When a WUC item operates in a different mode to perform each of more than one function, a different alpha designator is assigned for each operating mode.

C.3 FUNCTIONAL TABULATION

The "Flight Safety Functional Tabulation" sheet is used to code the safety model for keypunching. The sheets are coded as follows (refer to Figure C-1) for an example).

- a. Columns 1 through 3. Used to identify the aircraft represented by the model. For certain aircraft modeled under this contract more than one model - designation series MDS - was included. For instance, a single functional deck was created for four MDSs of the F-4 aircraft. Cards with "F4~~␣~~"* in columns 1-3 were common to all aircraft. For example,

*~~␣~~ = blank

when these cards are combined with those carrying "F4E" in columns 1-3, then it produces an F-4E FSPT model deck.

- b. Columns 4 through 31. Contain the title of the function or the WUC item.
- c. Columns 32 through 36. Contain the left-justified WUC number.
- d. Columns 37 and 38. Blank
- e. Columns 39 through 46. Contain the assigned alpha designator for the function and/or the WUC. Column 39 contains either an L or an R, or is blank. The L and R designate left and right for those instances when the function and/or WUC pertains to the left or right side of the aircraft.
- f. Columns 47 and 48. Blank.
- g. Columns 49 through 55. Normally left blank, but are used after a deck is operational to substitute the data on a card for that stored in the computer by punching the line record number in this field.
- h. Columns 56 through 63. Identify the dependent functions for either the function or specific WUCs being coded. Column 56 may contain L, R or blank for the same purpose as that of column 39.
- i. Column 64. Contains the alphanumeric code of the "provisory factor" applicable to the link value assigned.
- j. Columns 65 through 69. Contain the alpha designator of a function that is an alternate for the function being coded. (Column 65 is used for "L" or "R" as in Column 39.) The presence of the "alternate alpha" flags the importance of the link dependency as being affected by the success probability of the alternate function.
- k. Column 70. Contains the work unit code dependency value (1 = 0.10; 2 = 0.20; A = 1.0). This value is applicable to all flight phases.
- l. Column 71. Contains special instructions to the computer through the use of letters F, S, or being blank. Cards with an "S" or "blank" in column 71 are used in sensitivity computations. Cards with an "F" document a functional relationships which, although present in the system, would produce an erroneous sensitivity value when combined with other nonindependent paths (having the same function in common at some higher level). The "F" prevents the computer from including the link in the sensitivity calculations.
- m. Columns 72 through 80. Contain functional dependences for each of nine flight phases as described in Section 3.2.1 of the text. Coding is the same as for column 70.

C. 4 DIAGRAM CONSTRUCTION

The diagrams produced under the contract document the functional inter-relationship of the aircraft systems considered in the model. In the interest of extending the useful life of the diagrams, WUC items are not shown, thereby eliminating the necessity of updating the diagrams with each (and sometimes frequent) change to the WUC manual.

As discussed earlier in this report, the diagrams represent the hierarchal structure of the paths from which the sensitivity values are derived. The diagrams, although consistent with the system schematic and reliability block diagrams, are not equivalent due to this hierarchal method of documentation. In the actual system, signals and/or fluids pass from one component to the next and are thus documented in schematics; conversely, the hierarchal approach only identifies the components that must operate to achieve a given function, independent of the direction and/or sequence of signal flow. This approach directly addresses the system impact of a component failure without the necessity of identifying the intrasystem secondary failures. Each line connecting functions on the diagram is documented by a punchcard, with the lower function providing the "alpha designator" and the higher function's alpha designator indicator as the "dependent function". *

*The card deck also documents functional relationships not shown on the diagram; the work unit codes (mentioned earlier) and the "S" cards discussed in paragraph C. 3.1.

APPENDIX D
FSPT DOCUMENTATION OF T-39A AIRCRAFT

FSPT DOCUMENTATION OF T-39A AIRCRAFT

This appendix contains the functional relationship diagrams and a listing of the keypunch cards that comprise the T-39A aircraft FSPT safety model documentation.

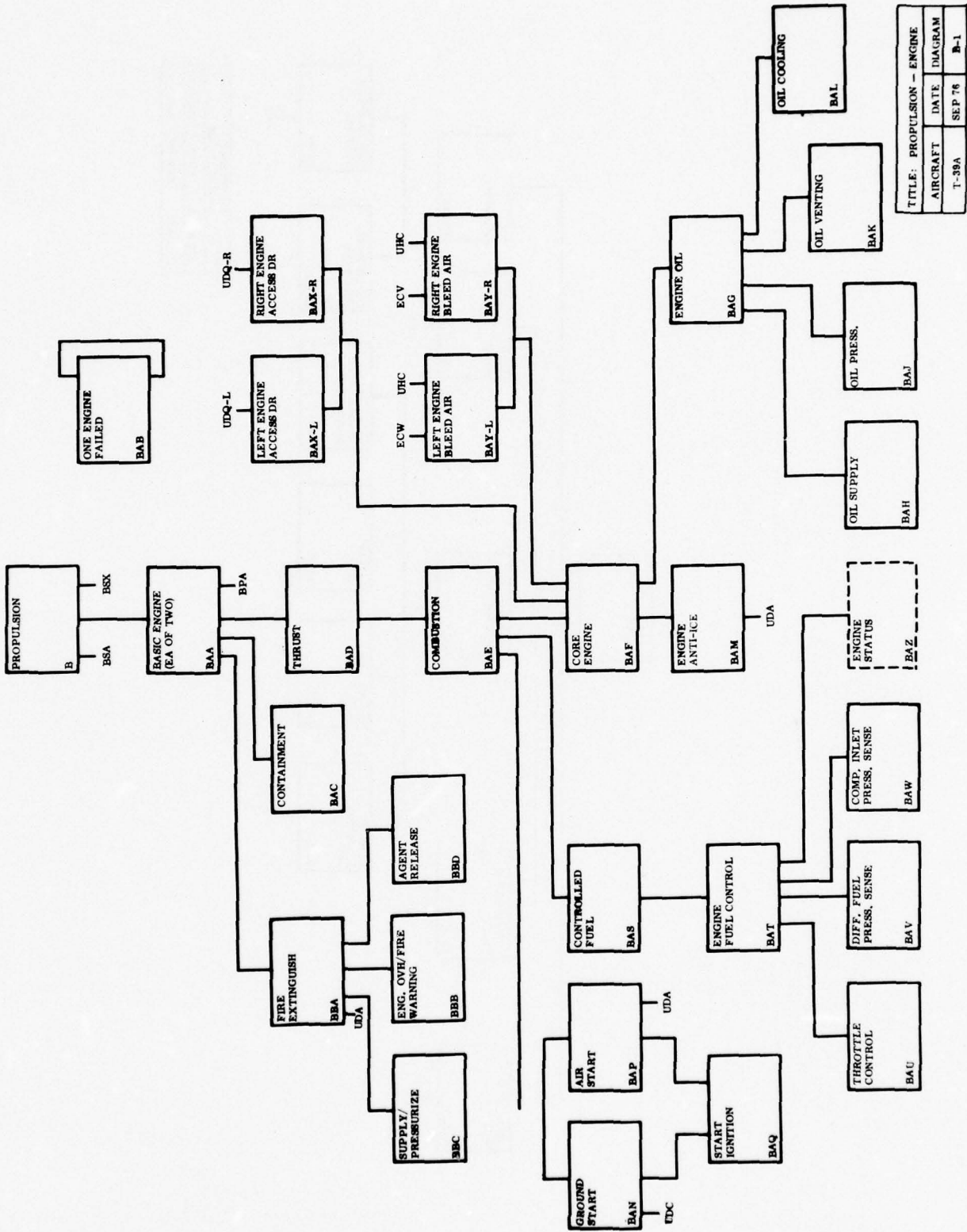
D.1 DIAGRAMS

The diagrams illustrating the functional relationships considered in the T-39A safety model will be found on pages D-5 through D-19, and are listed below:

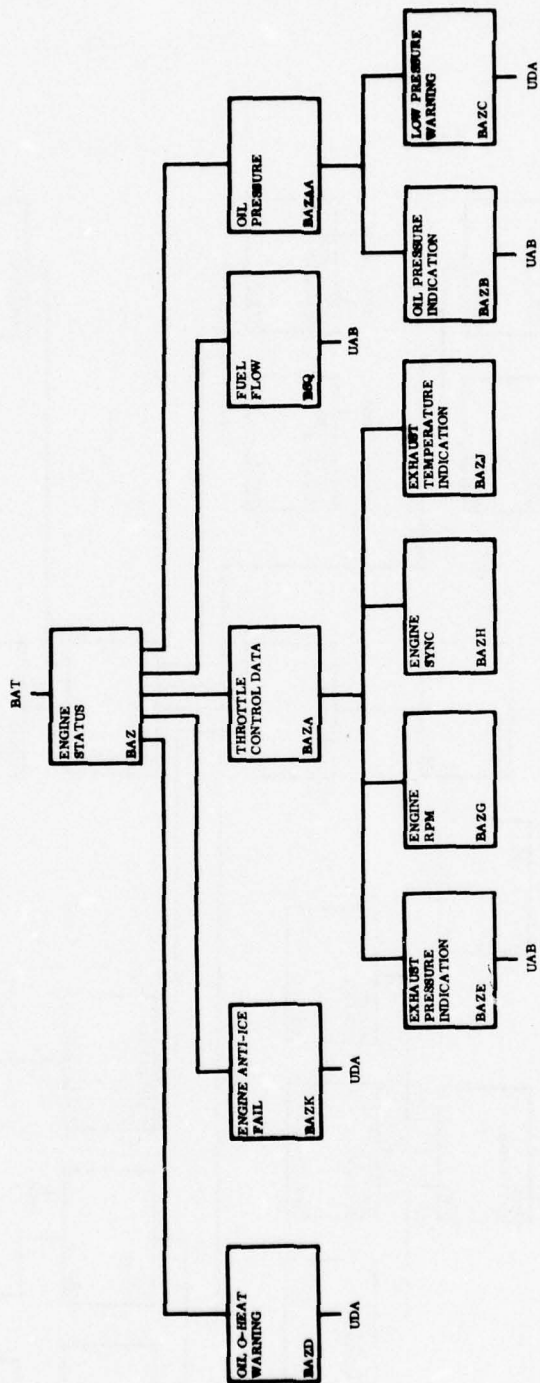
	<u>Diagram</u>	<u>Page</u>
Propulsion		
Propulsion/Engine	B-1	D-5
Propulsion/Engine Status	B-2	D-6
Propulsion/Fuel	B-3	D-7
Communications/Navigation/Identification		
Comm/Nav/Ident	C-1	D-8
En Route Aids	C-2	D-9
Information and Displays		
Info & Displays	D-1	D-10
Environmental Control		
Environmental Control	E-1	D-11
Environmental Control	E-2	D-12
Flight Control		
Flight Control	F-1	D-13
Yaw/Pitch Control	F-2	D-14
Ground Control	G-1	D-15
Mission Support	M-1	D-16
Landing Gear		
Landing Gear	N-1	D-17
Gear Retract	N-2	D-18
Utilities	U-1	D-19

D.2 CARD LISTING

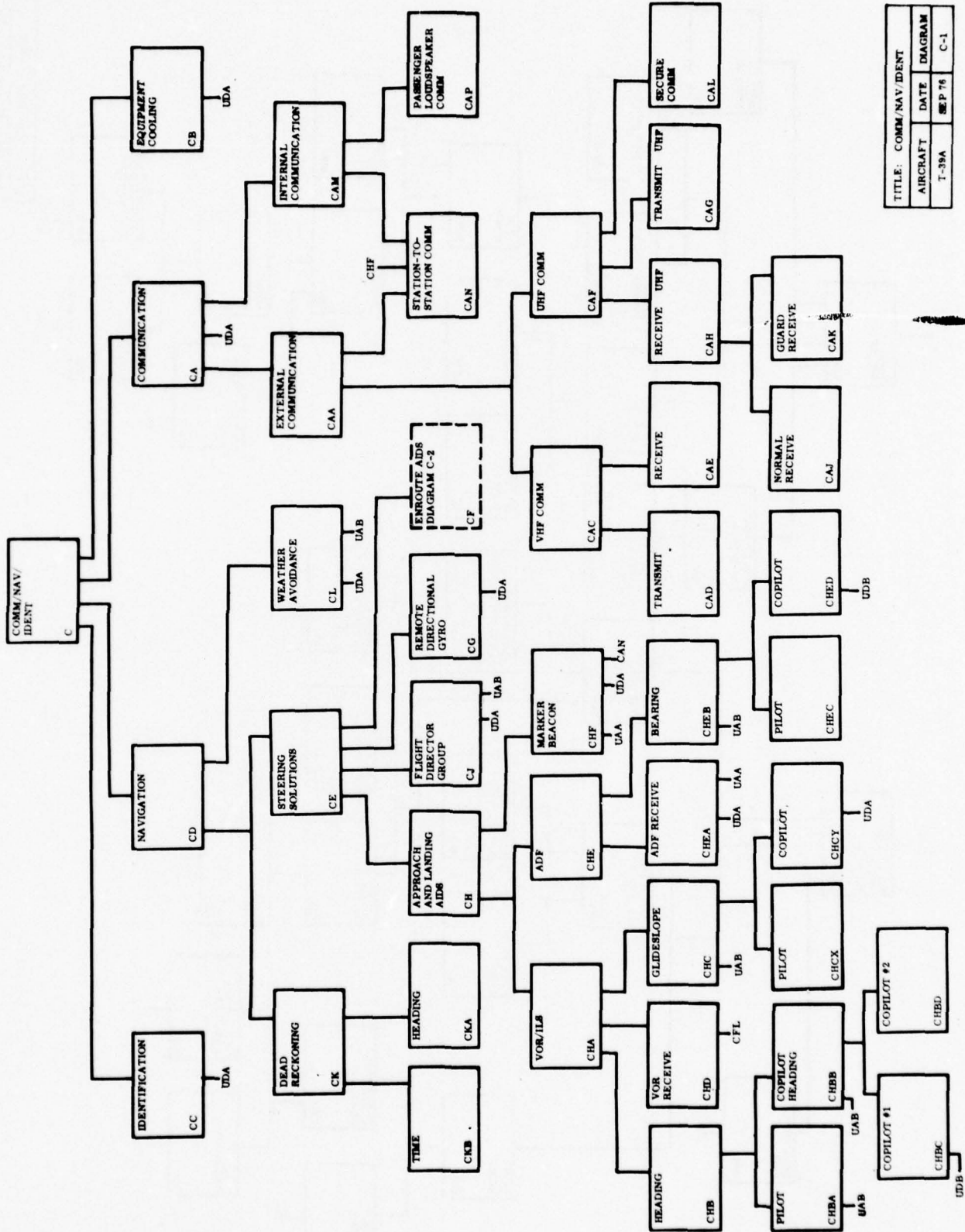
Pages D-21 through D-57 are a reproduction of the punchcard listing. The listing is alphabetical by "alpha designator", and the format is that of the 80-column punchcard itself as described in Appendix C. At the top of each page the card columns are printed vertically; for example, column 34 is printed "3".



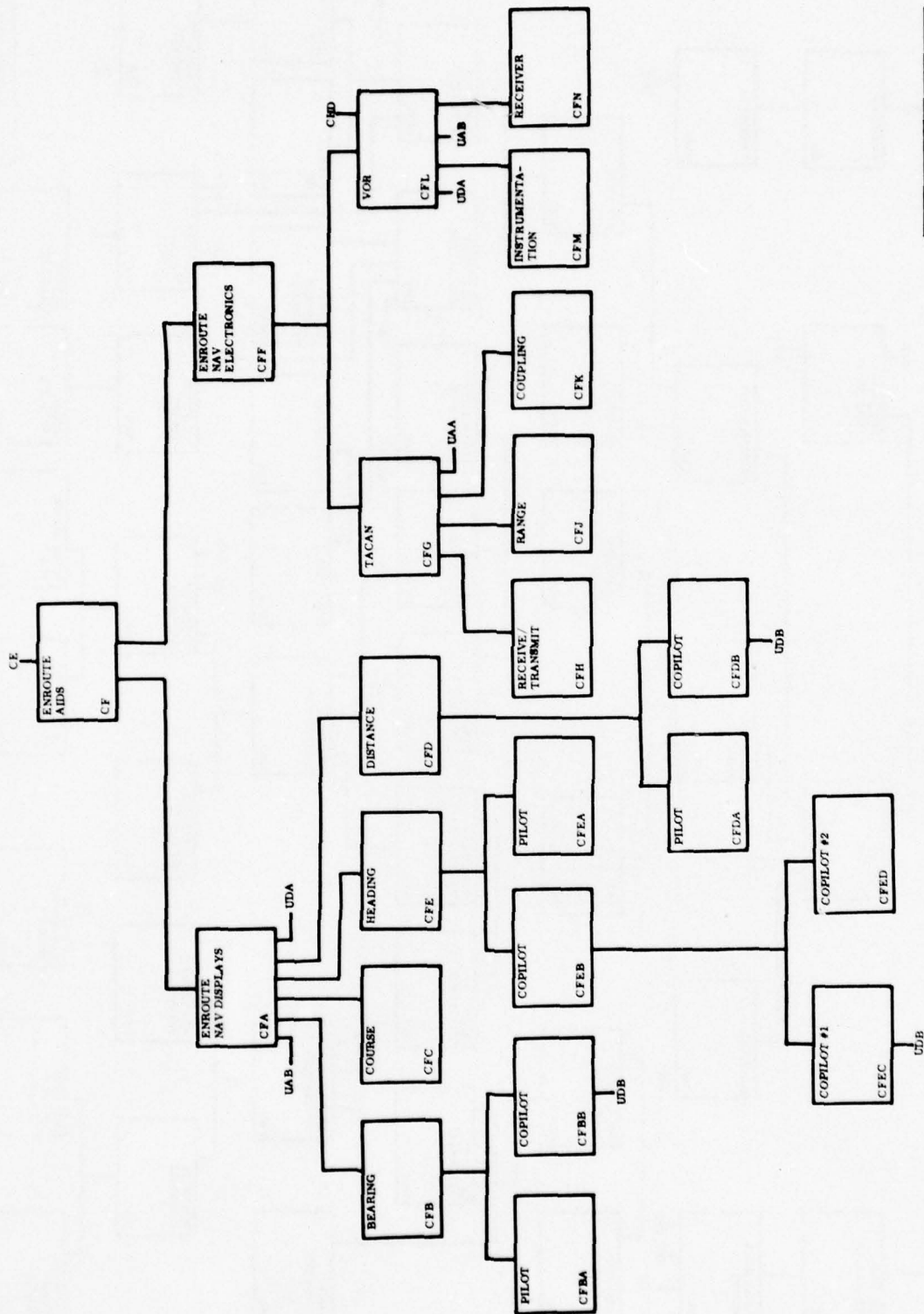
TITLE: PROPULSION - ENGINE			
AIRCRAFT	DATE	DIAGRAM	
T-39A	SEP 76	B-1	



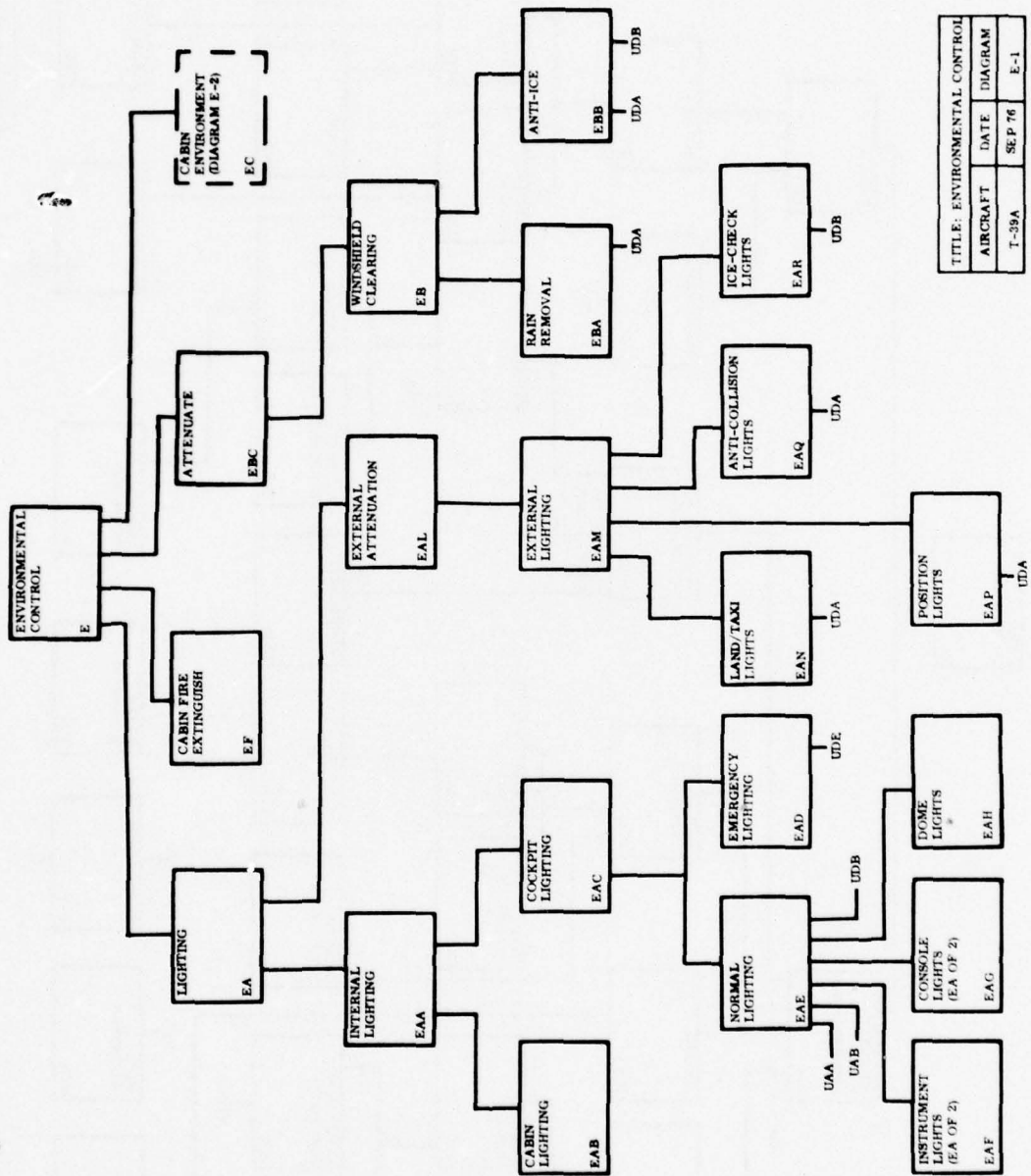
TITLE:	PROPULSION -
	ENGINE STATUS
AIRCRAFT:	DATE
T-38A	SEP 76
	DIAGRAM
	B-2



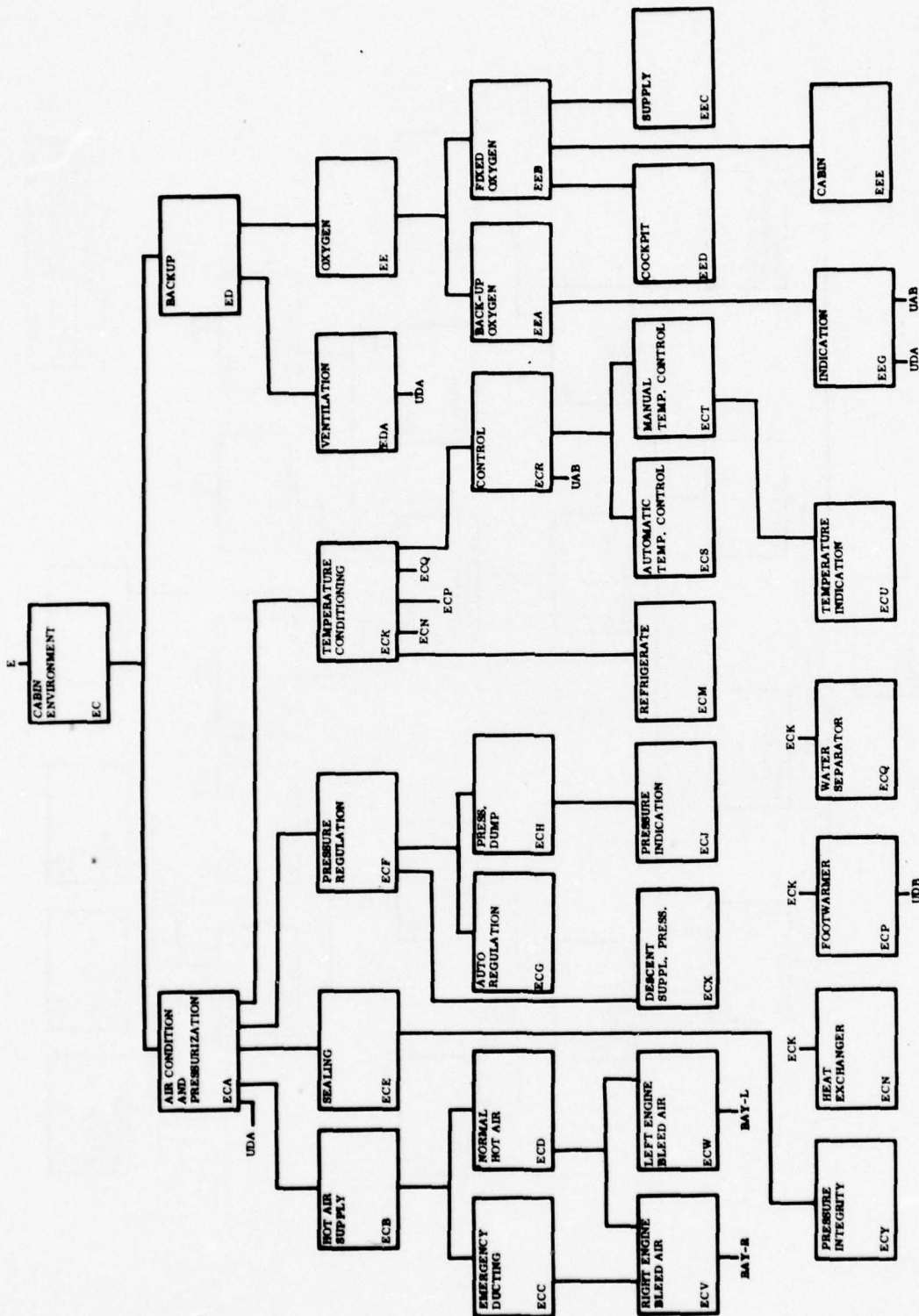
TITLE: COMM/NAV/IDENT		
AIRCRAFT	DATE	DIAGRAM
T-39A	SEP 76	C-1



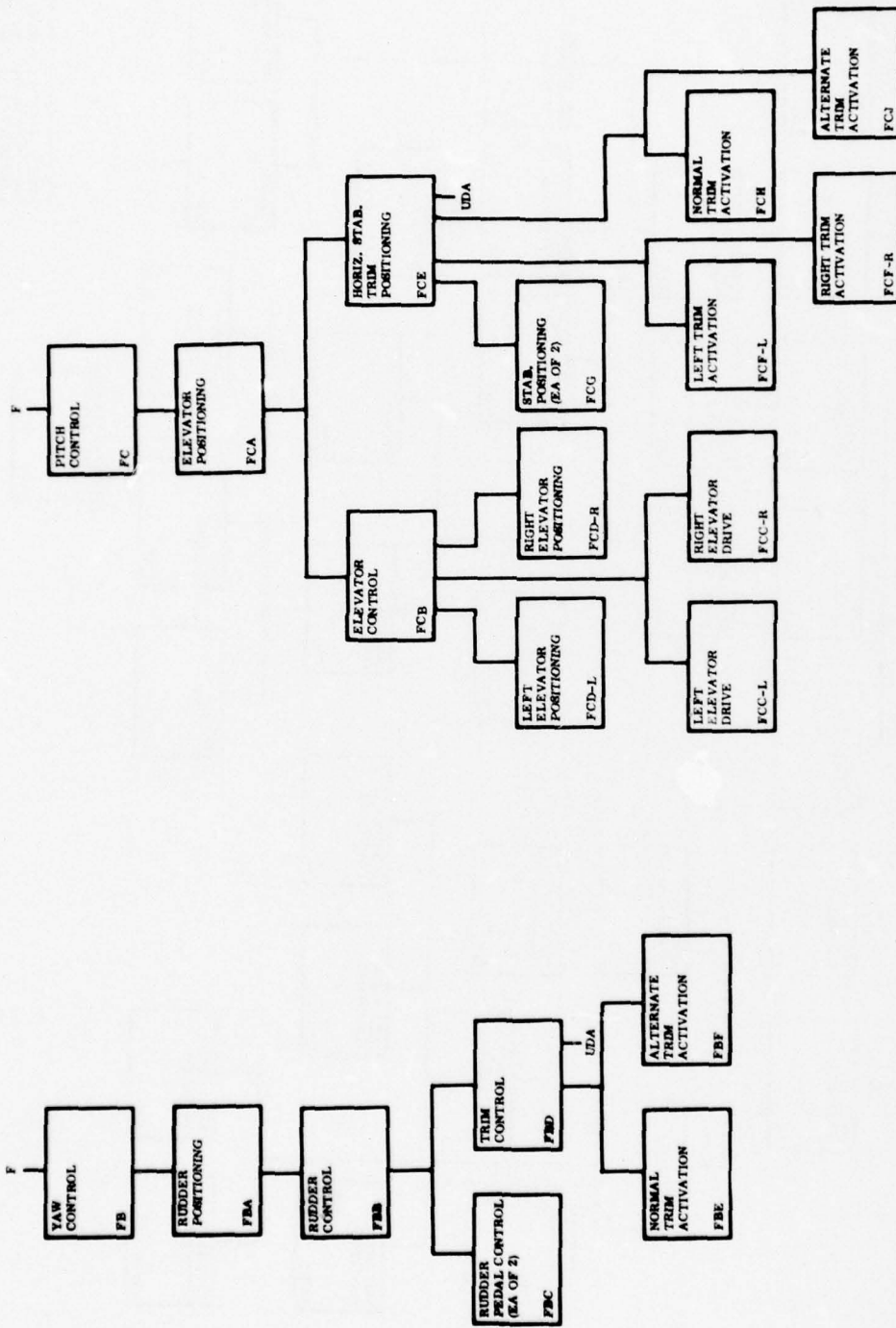
TITLE: COMM/NAV/IDENT			
AIRCRAFT	DATE	DIAGRAM	
T-39A	SEP 76	C-2	



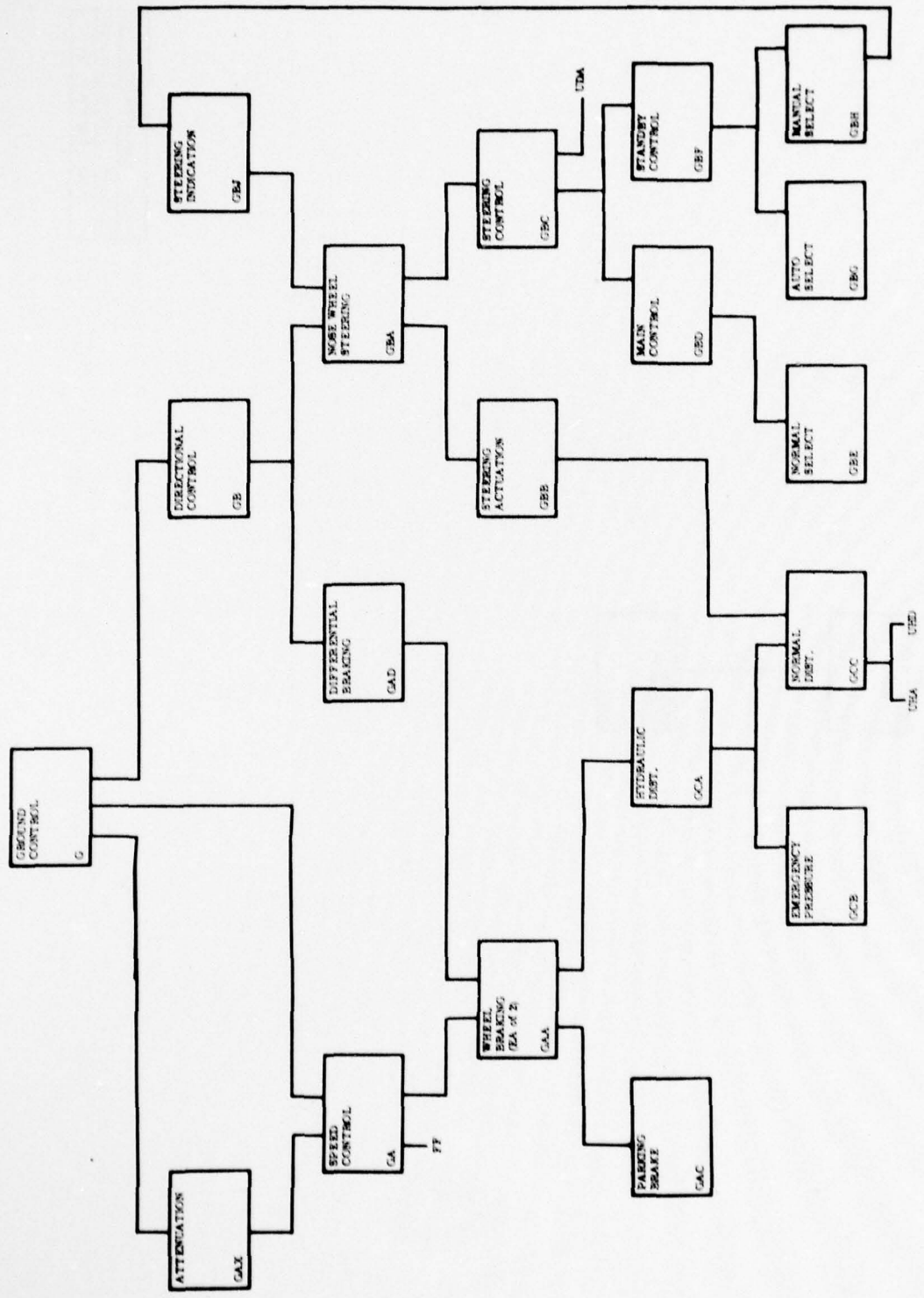
TITLE: ENVIRONMENTAL CONTROL		
AIRCRAFT	DATE	DIAGRAM
T-39A	SEP 76	E-1



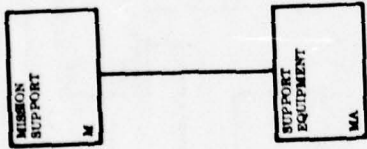
TITLE: ENVIRONMENTAL CONTROL			
AIRCRAFT	DATE	DIAGRAM	
T-39A	SEP 78	E-2	



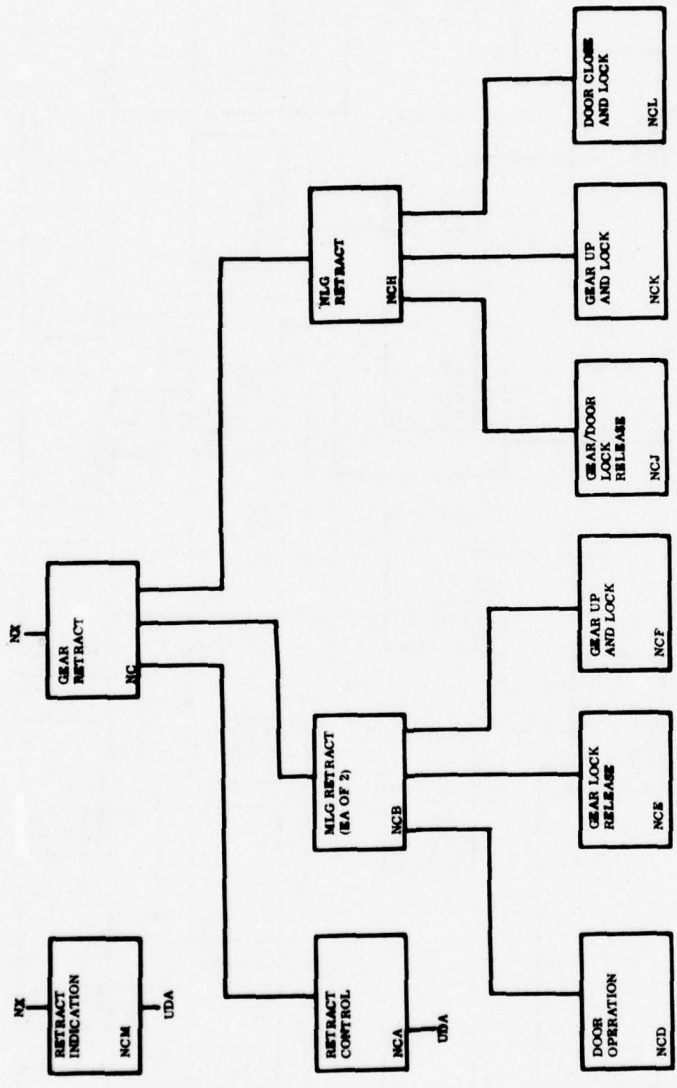
TITLE: FLIGHT CONTROL			
AIRCRAFT	DATE	DIAGRAM	
T-38A	SEP 76	F-2	



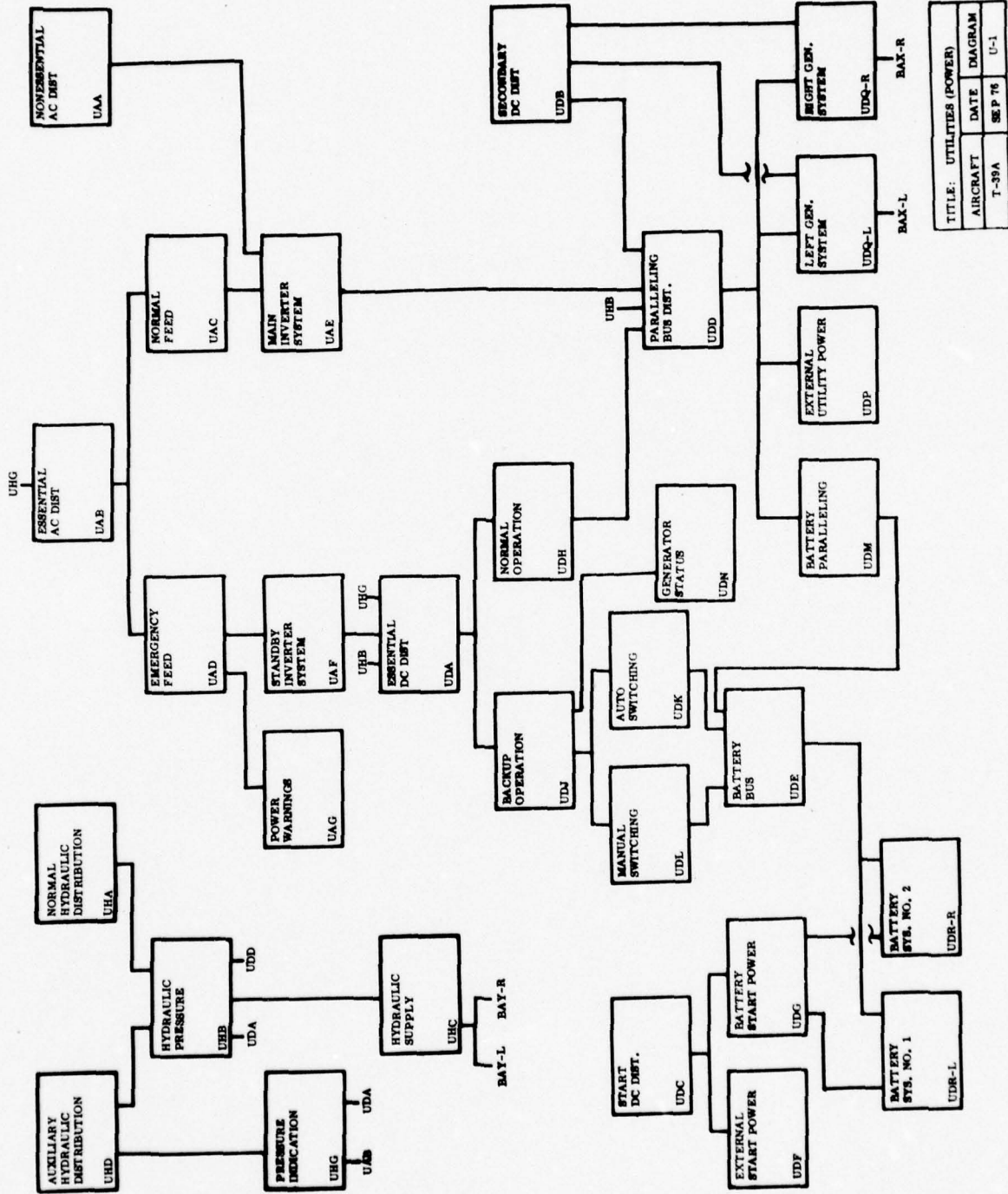
TITLE:	GROUND CONTROL
AIRCRAFT:	DATE
T-13A	DIAGRAM
REP	FIG
	G-1



TITLE: MISSION SUPPORT	
AIRCRAFT	DATE
T-39A	REPTS
	M-1



TITLE: LANDING GEAR			
AIRCRAFT	DATE	DIAGRAM	
T-39A	SEP 78	N-2	



TITLE: UTILITIES (POWER)		
AIRCRAFT	DATE	DIAGRAM
T-39A	SEP 76	U-1

BEST AVAILABLE COPY

PGG095.JIR1 DATE = 09/09/76

FLIGHT SAFETY PREDICTION TECHNIQUE

0000000001111111112222222222333333333344444444445555555555666666666677777777778
 12345678901234567890123456789012345678901234567890123456789012345678901234567890

PGG095.JICC T-39	T39		1 T39	
T39 PROPULSION		B		AAAAAAAAAA
T39 BASIC ENGINE, EA OF TWO		BAA	B	041111110
T39 BASIC ENGINE, EA OF TWO		BAA	B	OAAAAAAAAA
T39 BASIC ENGINE, EA OF TWO		BAA	BAB	SAAAAAAAAA
T39 ONE ENGINE FAILED		BAB	BAB	FAAAAAAAAA
T39 CONTAINMENT		BAC	BAA	AAAAAAAAAA
T39 FLEX MOUNT	2311A	BACA	BAC	A
T39 STEADY MOUNT	2311B	BACB	BAC	A
T39 DEFLECTOR	1161C	BACC	BAC	1
T39 TRUNNION MOUNT	1161H	BACD	BAC	1
T39 RETAINER	2311A	BACE	BAC	1
T39 SUPPORT	23115	BACF	BAC	1
T39 BRACKET	23116	BACG	BAC	1
T39 TRUNNION	23117	BACH	BAC	A
T39 TRUNNION SUPPORT	23118	BACJ	BAC	1
T39 SCOOP, RAM AIR	1161E	BACK	BAC	0
T39 DOOR, UPR, FWD, INBRD	11612	BACL	BAC	C
T39 DOOR, UPR, FWD, OUTBRD	11613	BACM	BAC	0
T39 DOOR, UPR, AFT	11614	BACN	BAC	0
T39 DOOR, LOWER	11615	BACP	BAC	0
T39 LATCH ASSY	11617	BACQ	BAC	0
T39 SKIN	11618	BACR	BAC	0
T39 SKIN	11632	BACS	BAC	0
T39 ACCESS DOOR	11633	BACT	BAC	0
T39 FRAME	11634	BACU	BAC	1
T39 THRUST		BAD	BAA	AAAAAAAAAA
T39 TAIL PIPE	23176	BADA	BAD	1
T39 TAIL PIPE CLAMP	23177	BADB	BAD	2
T39 COMBUSTION		BAE	BAD	AAAAAAAAAA
T39 CASE ASSY, COMB CHB, INNER	23AEA	BAEA	BAE	A
T39 CBR ASSY, COMB NO 1,5,7	23AEB	BAEB	BAE	A
T39 CBR ASSY, COMB NO 2,4,8	23AEC	BAEC	BAE	A
T39 CHB ASSY, COMB NO 3	23AED	BAED	BAE	A
T39 CHB ASSY, COMB NO 6	23AEE	BAEE	BAE	A
T39 CLAMP ASSY, CHB RETAINING	23AEF	BAEF	BAE	A
T39 CASE ASSY, COMB CHB, OUTER	23AEG	BAEG	BAE	A
T39 NUT ASSY, COMB CHB FIRESEAL	23AEH	BAEH	BAE	A
T39 FUEL DRAIN VALVE	2315C	BAEJ	BAE	1
T39 DUCT ASSY, COMB CHB OUTLET	23AFA	BAEK	BAE	5
T39 SUP ASSY, CHB OUTLET DUCT	23AFC	BAEL	BAE	5
T39 CORE ENGINE		BAF	BAE	AAAAAAAAAA
T39 CORE ENGINE		BAF	LBAX	FAAAAAAAAA
T39 CORE ENGINE		BAF	RBAX	FAAAAAAAAA
T39 CORE ENGINE		BAF	LBAY	FAAAAAAAAA
T39 CORE ENGINE		BAF	RBAY	FAAAAAAAAA
T39 COMP ROTOR/STATOR ASSY	23AAA	BAFA	BAF	8
T39 COMP INLET CASE	23AAE	BAFB	BAF	8
T39 BRG NO 1	23AAG	BAFC	BAF	A
T39 SEAL, NO 1 BRG	23AAH	BAFD	BAF	8

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T39 HOUSING, NO 1 BRG	23AAJ	BAFE	BAF	1
T39 VANE, COMP INLET	23AAL	BAFF	BAF	A
T39 VANE/SHROUD ASSY-1ST STA	23AAM	BAFG	BAF	8
T39 VANE/SHROUD ASSY-2ND STA	23AAN	BAFH	BAF	8
T39 VANE/SHROUD ASSY-3RD STA	23AAP	BAFJ	BAF	8
T39 PLATE, COMP VANE, INNER	23AAW	BAFK	BAF	A
T39 VANE, 1ST STA	23AAZ	BAFL	BAF	A
T39 FRONT HUB	23AAB	BAFM	BAF	8
T39 BLADE, 1ST STA	23ACD	BAFN	BAF	A
T39 BRG, COMP, REAR	23AC2	BAFQ	BAF	A
T39 SEAL ASSY, COMP RR BRG HSN	23AC8	BAFR	BAF	8
T39 DIFFUSER CASE	23ADA	BAFS	BAF	A
T39 VANE AND SHROUD ASSY	23ADD	BAFT	BAF	8
T39 SHROUD ASSY, 1ST STA VANE	23AFB	BAFU	BAF	8
T39 CASE ASSY, TURBINE	23AFD	BAFV	BAF	8
T39 SEAL, TB ROTOR, 1STA, OUTER	23AFE	BAFW	BAF	8
T39 SPACER, TB ROTOR, 1STA SEAL	23AFG	BAFX	BAF	1
T39 SHROUD/SEAL ASSY, TB VANE	23AFH	BAFY	BAF	1
T39 SEAL, TB ROTOR, 2ND STA	23AFJ	BAFZ	BAF	8
T39 VANE, TB, 1ST STA	23AFM	BAFZA	BAF	A
T39 VANE, TB, 2ND STA	23AFN	BAFZB	BAF	A
T39 TURBINE ROTOR	23AFP	BAFZC	BAF	A
T39 SEAL ASSY, TB BRG	23AFQ	BAFZD	BAF	8
T39 SUPPORT, TB BRG SEAL	23AFS	BAFZE	BAF	1
T39 SHIELD ASSY, TB BRG, HEAT	23AFT	BAFZF	BAF	2
T39 DISK, 2ND STA TB	23AFW	BAFZG	BAF	8
T39 RING, RETAINING BRG IN RACE	23AF2	BAFZH	BAF	8
T39 BRG, TURBINE	23AF3	BAFZJ	BAF	A
T39 BLADE, TB, 1ST STA	23AF5	BAFZK	BAF	A
T39 BLADE, TB, 2ND STA	23AF6	BAFZL	BAF	A
T39 CASE ASSY, TB EXHAUST	23AF7	BAFZM	BAF	A
T39 CONE AND STRUT ASSY	23AF8	BAFZN	BAF	8
T39 NOSE GUIDE VANE	1161G	BAFZP	BAF	8
T39 AIR INLET DUCT NOSE ASSY	<11616	BAFZQ	BAF	2
T39 ENGINE OIL DISTRIBUTION		BAG	BAF	00AAAAA00
T39 TUBE-TURBINE BRG TRANSFER	23CAP	BAGA	BAG	8
T39 OIL SUPPLY		BAH	BAG	AAAAA0000
T39 MAIN OIL STRAINER	23CAB	BAHA	BAH	2
T39 SPT ASSY, OIL STRAINER	23CAC	BAHB	BAH	1
T39 OIL TANK ASSY	23CAE	BAHC	BAH	8
T39 TUBE ASSY, FR BRG-PUMP RTN	23CAH	BAHD	BAH	1
T39 TUBE ASSY, TB BRG EXT RTN	23CAJ	BAHE	BAH	1
T39 STRAINER ASSY, CMP RR BRG	23CAM	BAHF	BAH	1
T39 STRAINER ASSY, TB BRG	23CAN	BAHG	BAH	1
T39 STRAINER SCREEN GEARBX OIL	23AHP	BAHH	BAH	1
T39 OIL PRESSURIZATION		BAJ	BAG	AAAAA0000
T39 MAIN OIL PUMP	23CAA	BAJA	BAJ	A
T39 OIL PRESS RELIEF VALVE	23CAD	BAJB	BAJ	2
T39 MANF ASSY, PRESS MAIN BRG	23CAK	BAJC	BAJ	8
T39 NOZZLE ASSY, CMP RR BRG	23CAL	BAJD	BAJ	1

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T39	NOZZLE CMP FR BRG	23CAR	BAJE	HAJ					1
T39	SEAL, STARTER/GEN PAD	23AJ5	HAJF	HAJ					1
T39	TUBE ASSY FRONT BEAR PRES	23CAG	BAJG	HAJ					1
T39	VENTING		BAK	BAG					00000000
T39	MANF ASSY, TANK BREATHER	23CBB	BAKA	BAK					1
T39	MANF ASSY, BREATHER, EXT FR	23CBC	BAKB	BAK					1
T39	TUBE ASSY, CMP FR BRG BRTHR	23CBD	BAKC	BAK					1
T39	SEAL BREATHER SHAFT GEAR	23AJG	BAKD	BAK					0
T39	OIL COOLING		BAL	BAG					004888000
T39	FUEL OIL COOLER	23517	BALA	HAL					8
T39	ENGINE ANTI-ICE		BAM	BAF	A				888888888
T39	RELAY	23413	BAMA	BAM					A
T39	ENG INLET DE-ICE VALVE	23417	BAMB	BAM					A
T39	DUCT	23418	BAMC	BAM					A
T39	CIRCUIT BREAKER	42152	BAMD	BAM					1
T39	GROUND START		BAN	BAE					00000000
T39	HSNG, START/GEN DR SHT BRG	23AHL	BANA	BAN					1
T39	SHAFT, STARTER/GEN DR	23AJ3	BANB	BAN					A
T39	SUPPORT ASSY, START/GEN BRG	23AJ7	BANC	BAN					1
T39	COVER, START/GEN ACCESS DR	23AJ8	BAND	BAN					0
T39	STARTER BUTTON	23312	BANE	BAN					A
T39	STARTER-GENERATOR	42131	BANF	HAN					A
T39	RLY, GENERATOR, MONITOR	4217C	BANG	BAN					A
T39	RLY, GENERATOR, FIELD CONTR	4217D	BANH	BAN					A
T39	RLY, BATTERY START	42171	BANJ	BAN					A
T39	RLY, EXT PWR CUTOFF	42172	BANK	HAN					0
T39	RELAY START CONTROLLER	4217J	HANL	BAN					A
T39	CIRCUIT BREAKER	42152	HANM	BAN					1
T39	AIR START		BAP	BAE	T				00AAAAA00
T39	SW., AIR START	23313	BAPA	BAP					A
T39	RELAY	23314	BAPB	BAP					A
T39	CIRCUIT BREAKER	42152	BAPC	BAP					1
T39	START/IGNITION		BAQ	BAN					00000000
T39	START IGNITION		BAQ	BAP					0AAAAAA00
T39	EXCITER, IGNITION	23FAB	BAQA	HAQ					A
T39	CABLE, ELEC PWR, EXC-IGN	23FAC	BAQB	HAQ					A
T39	IGNITER PLUG #4EAC	23FAD	BAQC	HAQ					2
T39	CABLE, BROND, AIRFRAME/ENG	23FAE	BAQD	HAQ					A
T39	SW., ENGINE MASTER	23131	BAQE	BAQ					A
T39	SW., FUEL-IGNITION	23311	BAQF	BAQ					A
T39	LIGHT INDICATOR	23318	BAQG	RAQ					0
T39	CIRCUIT BREAKER	23132	BAQH	BAQ					A
T39	CABLE ASSY PN 10 166435	23123	BAQJ	RAQ					8
T39	CONTROLLED FUEL		BAS	BAE					AAAAA0000
T39	MANIFOLD ASSY, FUEL	23BAK	BASA	BAS					A
T39	NOZZLE ASSY, FUEL METERING	23BAN	BASB	BAS					2
T39	CONNECTOR, PRESS VALVE UNK	23BAP	BASC	BAS					0
T39	TUBE, XFER, PRESS VALVE	23BAQ	BASD	BAS					2
T39	TUBE ASSY, FUEL SIGNAL	23BAT	BASE	BAS					A
T39	DRAIN MANIFOLD	2311F	BASF	RAS					0

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T39 DIAPHRAGM 23112 BASG BAS 3
T39 DIAPHRAGM SEAL 23113 BASH BAS 3
T39 FUEL PRESS & DUMP VALVE 23BAR BASJ BAS 8
T39 VALVE ASSY,FUEL CHECK 23BAS BASK BAS 0
T39 ENGINE FUEL CONTROL 23BAJ BAT BAS AAAAAAAA
T39 FUEL CONTROL 23BAC BATA BAT 5
T39 FILTER SCREEN & SPRT ASSY 23BAD BATB BAT 1
T39 BLEED ACTUATOR ASSY 23BAE BATC BAT 3
T39 POWER LEVER X-SHAFT ASSY 23BAF BATD BAT 8
T39 SUPPORT, X-SHAFT 23BAG BATE BAT 1
T39 ARM/LEVER, X-SHAFT 23BAJ BATF BAT A
T39 GEAR SHAFT, FUEL CONTR DR 23AJC BATJ BAT A
T39 BOSS ASSY, FUEL CONTROL 23AJE BATK BAT 5
T39 THROTTLE CONTROL 23221 BAU BAT AAAAAAAA
T39 QUADRANT ASSY 23221 BAUA BAU 8
T39 FRICTION LOCK 23222 BAUB BAU 1
T39 ROD ASSY 23226 BAUC BAU 8
T39 ROD ASSY, LH 23227 BAUD BAU 8
T39 ROD ASSY, RH 23228 BAUE BAU 8
T39 TELESCOPIC UNIT 2322C BAUF BAU 8
T39 TELEFLEX CABLE 2322A BAUG BAU A
T39 DIFFUSER PRESS SENSE 23ADJ BAV BAT AAAAAAAA
T39 PROBE, FUEL CONT PRESS SENS 23ADJ BAVA BAV A
T39 TUBE ASSY, DIFF PRESS SENS 23BAU BAVB BAV A
T39 COMP INLET PRESS SENSE 23AAK BAW BAT AAAAAAAA
T39 PROBE ASSY, COMP IN PRESS 23AAK BAWA BAW A
T39 TUBE ASSY, COMP IN PRESS 23BAV BAWB BAW 8
T39 LEFT ENG ACCESS DR LBAX RPF8 SAAAAAAAAA
T39 LEFT ENG ACCESS DR LBAX LUDQ AAAAAAAA
T39 RIGHT ENG ACCESS DR RBAX RUDQ AAAAAAAA
T39 CPL, GEAR BOX DR SHAFT 23AGB LBAXA LBAX A
T39 CPL, GEAR BOX DR SHAFT 23AGB RBAXA RBAX A
T39 SPACER, UPR BRG SLEEVE, IN 23AGD LBAXB LBAX 1
T39 SPACER, UPR BRG SLEEVE, IN 23AGD RBAXB RBAX 1
T39 GEAR BOX 23AHA LBAXC LBAX 8
T39 GEAR BOX 23AHA RBAXC RBAX 8
T39 PLUG, DR SHAFT 23AHC LBAXD LBAX 0
T39 PLUG, DR SHAFT 23AHC RBAXD RBAX 0
T39 ADPTR, GEAR BOX-POSN BOSS 23AHD LBAXE LBAX 0
T39 ADPTR, GEAR BOX-POSN BOSS 23AHD RBAXE RBAX 0
T39 HSG ASSY, DR GEARBOX 23AHG LBAXF LBAX 1
T39 HSG ASSY, DR GEARBOX 23AHG RBAXF RBAX 1
T39 COVER, ENG ACCESS DR 23AJN LBAXG LBAX 1
T39 COVER, ENG ACCESS DR 23AJN RBAXG RBAX 1
T39 GEAR MAIN COMP DRIVE 23ACW LBAXH LBAX A
T39 GEAR MAIN COMP DRIVE 23ACW RBAXH RBAX A
T39 LEFT ENG BLEED AIR DISTR LBAY ECW AAAAAAAA
T39 LEFT ENG BLEED AIR DISTR LBAY UHC 11111111
T39 RIGHT ENG BLEED AIR DISTR RBAY ECV AAAAAAAA
T39 RIGHT ENG BLEED AIR DISTR RBAY UMC 11111111

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T39 BLEED VALVE & LINKAGE      23DAA LBAYA      LBAY      A
T39 BLEED VALVE & LINKAGE      23DAA RBAYA      RBAY      A
T39 STRAP ASSY,COMPRESSOR      23DAB LBAYB      LBAY      8
T39 STRAP ASSY,COMPRESSOR      23DAB RBAYB      RBAY      8
T39 CONNECT LINK,BLEED VALVE   23DAC LBAYC      LBAY      A
T39 CONNECT LINK,BLEED VALVE   23DAC RBAYC      RBAY      A
T39 ARM,AIR BLEED VALVE        23DAH LBAYD      LBAY      A
T39 ARM,AIR BLEED VALVE        23DAH RBAYD      RBAY      A
T39 ROD,VALVE LINKAGE          23DAL LBAYE      LBAY      A
T39 ROD,VALVE LINKAGE          23DAL RBAYE      RBAY      A
T39 ENGINE STATUS              BAZ      BAT      AAAAAA4AA
T39 THROTTLE CONTROL DATA ATTN BAZ      BAZ      111111111
T39 OIL PRESSURE                BAZAA    BAZ      I BAG  AAAAAA8AA
T39 OIL PRESS INDICATION        BAZB     BAZAA    111111111
T39 INDICATOR,OIL PRESSURE      51441    BAZBA    BAZB     A
T39 XMITTER,OIL PRESSURE        51442    BAZBB    BAZB     A
T39 CIRCUIT BREAKER            42231    BAZBC    BAZB     1
T39 LOW PRESS WARNING          BAZC     BAZAA    111111111
T39 SW.,OIL PRESSURE           51443    BAZCA    BAZC     A
T39 LIGHT PANEL                44241    BAZCB    BAZC     1
T39 LIGHT,MASTER              44242    BAZCC    BAZC     1
T39 LIGHT,INDICATOR            44243    BAZCD    BAZC     2
T39 RELAY,TEST                 44244    BAZCE    BAZC     0
T39 CIRCUIT BREAKER            42152    BAZCF    BAZC     1
T39 OIL OVERHEAT WARNING        BAZD     BAZ      I BAL  111111111
T39 BULB,OIL TEMP              51426    BAZDA    BAZD     A
T39 LIGHT PANEL                44241    BAZDB    BAZD     1
T39 LIGHT,MASTER              44242    BAZDC    BAZD     1
T39 LIGHT,INDICATOR            44243    BAZDD    BAZD     2
T39 RELAY,TEST                 44244    BAZDE    BAZD     0
T39 CIRCUIT BREAKER            42152    BAZDF    BAZD     1
T39 EXHAUST PRESS IND          BAZE     BAZA     032222222
T39 INDICATOR,EXHAUST PRESS    51431    BAZEA    BAZE     A
T39 XMITTER,EXHAUST PRESS      51432    BAZEB    BAZE     4
T39 MANIFOLD ASSY,UPPER LEFT    23HAA    BAZEC    BAZE     5
T39 MANIFOLD ASSY,LOWER LEFT   23HAB    BAZED    BAZE     5
T39 MANIFOLD ASSY,LOWER RIGHT  23HAC    BAZEE    BAZE     5
T39 MANIFOLD ASSY,UPPER RIGHT  23HAD    BAZEF    BAZE     5
T39 CIRCUIT BREAKER            42231    BAZEG    BAZE     1
T39 ENGINE RPM                  BAZG     BAZA     111111111
T39 INDICATOR,TACHOMETER       51411    BAZGA    BAZG     A
T39 GENERATOR,TACHOMETER       51412    BAZGB    BAZG     A
T39 GEARSHIFT TACH DRIVE       23AJR    BAZGC    BAZG     8
T39 SEAL TACH DRIVE            23AJV    BAZGD    BAZG     1
T39 ENGINE SYNC                BAZH     BAZA     000000000
T39 SYNCHROSCOPE              51413    BAZHA    BAZH     A
T39 EXHAUST GAS TEMP INDICATION BAZJ     BAZA     111111111
T39 INDICATOR,EXHAUST PRESS    51421    BAZJA    BAZJ     A
T39 EXHAUST TEMP RES SPOOL     51422    BAZJB    BAZJ     A
T39 THERMOCOUPLE              23GAA    BAZJC    BAZJ     A
    
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T39 CABLE ASSY, THERMOCOUPLE 23GAH BAZJD BAZJ 1
T39 ENGINE ANTI-ICE FAIL BAZK BAZ K BAM 333333333
T39 RELAY 23413 BAZKA BAZK A
T39 LIGHT PANEL 44241 BAZKB BAZK 1
T39 LIGHT MASTER 44242 BAZKC BAZK 1
T39 LIGHT INDICATOR 44243 BAZKD BAZK 2
T39 RELAY, TEST 44244 BAZKE BAZK 0
T39 CIRCUIT BREAKER 42152 BAZKF BAZK 1
T39 ENGINE FIRE EXTINGUISH BBA BAA X AAAAAAAAAA
T39 SHUT-OFF VALVE 46222 BBAA BBA A
T39 SW., ENG MASTER 23131 BBAB BBA A
T39 CABLE ASSY PN 10 166435 23133 BBAC BBA 8
T39 CIRCUIT BREAKER 42152 BBAD BBA 1
T39 ENGINE OVHT/FIRE WARNING BBB BBA AAAAAAAAAA
T39 DETECTOR ELEMENT 49111 BBBA BBB A
T39 INDICATOR LIGHT 49112 BBBD BBB A
T39 CONTROL UNIT 49113 BBBC BBB A
T39 QUICK DISCONNECT 49114 BBBB BBB 0
T39 LIGHT, PANEL 44241 BBBE BBB 1
T39 LIGHT, MASTER 44242 BBBF BBB 1
T39 LIGHT, INDICATOR 44243 BBBG BBB 2
T39 RELAY, TEST 44244 BBBH BBB 0
T39 SUPPLY & PRESSURIZATION BBC BBA AAAAAAAAAA
T39 CONTAINER ASSY XEA OF TWO 49211 BBBC BBC 5
T39 INDICATOR, DISCHARGE 49215 BBCE BBC 0
T39 SW., SELECTOR 49218 BBCC BBC A
T39 VALVE, CHECK 49214 BBCE BBC 0
T39 AGENT RELEASE BBD BBA AAAAAAAAAA
T39 VALVE, DISCHARGE 49212 BBDA BBD A
T39 HANDLE ASSY 49217 BBDB BBD A
T39 CARTRIDGE, FIRE EXTINGUISH 97355 BBDC BBD A
T39 ONE ENGINE FUEL PRESSURE BPA BAA AAAAAAAAAA
T39 TANK BOOST PRESSURE BPH BPA K BPE AAAAAAAAAA
T39 SHUT-OFF VALVE 46222 BPHA BPH 0
T39 ENGINE MASTER SWITCH 23131 BPBB BPH 0
T39 CIRCUIT BREAKER 23132 BPHC BPH 0
T39 CABLE ASSY PN 10 166435 23133 BPHD BPH 8
T39 SAME SIDE XNORMALC BPC BPH 111111111
T39 BOOST PUMP 46221 BPCA BPC A
T39 RELAY, BOOST 46227 BPCB BPC A
T39 MOTOR, BOOST PUMP 46228 RPCC BPC A
T39 OTHER SIDE BPD BPH K BPC AAAAAAAAAA
T39 BOOST PUMP 46221 BPDA BPD A
T39 RELAY, BOOST 46227 BPDF BPD A
T39 MOTOR, BOOST PUMP 46228 BPDF BPD A
T39 CROSS-FEED VALVE XBOOSTC 46224 BPDF BPD A
T39 FIRST STAGE ENG. BOOST PRES. BPE BPA K BPH AAAAAAAAAA
T39 PUMP, FUEL, ENG DRIVEN 23212 RPEA BPE A
T39 CIRCUIT BREAKER 42152 RPEB BPF 1
T39 ENGINE BOOST X2ND STAC BPF BPA AAAAAAAAAA
    
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T39	PUMP, FUEL, ENG DR	23212	BPFA	BPF	A
T39	ACCESSORY DRIVE, TYPICAL		BPFB	BPF	AAAAAAAAA
T39	FUEL FILTERING		BPG	BPF	111111111
T39	FILTER ELEMENT	23214	BPGA	BPG	A
T39	FUEL HEAT		BPH	BPG	Y AAAAAAAAA
T39	HEATER	23215	BPHA	BPH	A
T39	AIR VALVE AND TUBE	23FAA	BPHB	BPH	A
T39	VALVE/ACT ASSY, AIR SHUTOFF	23FAP	BPHC	BPH	A
T39	LIGHT, PANEL	44241	BPHD	BPH	0
T39	LIGHT, MASTER	44242	BPHE	BPH	0
T39	LIGHT, INDICATOR	44243	BPHF	BPH	0
T39	CIRCUIT BREAKER	42152	BPHG	BPH	1
T39	FILTER BYPASS		BPJ	BPF	K BPG AAAAAAAAA
T39	FILTER XBY-PASS VALVE	23213	BPJA	BPJ	A
T39	FUEL SUPPLY		BSA	B	0AAAAAAAA50
T39	LEFT TANK SUPPLY		LBSB	BSA	KRBSB AAAAAAAAA
T39	RIGHT TANK SUPPLY		RBSR	BSA	KLBSB AAAAAAAAA
T39	FUEL CELL, WING INTEGRAL	46110	LBSBA	LBSB	1
T39	FUEL CELL, WING INTEGRAL	46110	RBSBA	RBSB	1
T39	SEALANT	46115	LBSBB	LBSB	1
T39	SEALANT	46115	RBSBB	RBSB	1
T39	STAT-O-SEAL	46116	LBSBC	LBSB	1
T39	STAT-O-SEAL	46116	RBSBC	RBSB	1
T39	EJECTOR	46211	LBSBD	LBSB	0
T39	EJECTOR	46211	RBSBD	RBSB	0
T39	DRAIN VALVE	46212	LBSBE	LBSB	0
T39	DRAIN VALVE	46212	RBSBE	RBSB	0
T39	COUPLING	46213	LBSBF	LBSB	1
T39	COUPLING	46213	RBSBF	RBSB	1
T39	DRAIN VALVE	46113	LBSBG	LBSB	0
T39	DRAIN VALVE	46113	RBSBG	RBSB	0
T39	ATTEN, INT GRAV REPLENISH		BSC	BSA	111111111
T39	INTERNAL GRAV REPLENISH		BSD	BSC	000111000
T39	CELL ASSY	46121	BSDA	BSD	1
T39	SEAL	46124	BSDR	BSD	1
T39	DRAIN VALVE	46127	BSDC	BSD	0
T39	DRAIN LINE	46128	BSDD	BSD	0
T39	CHECK VALVE	46414	BSD E	BSD	0
T39	FUEL DUMP		BSE	FE	111111111
T39	NOZZLE	46411	BSEA	BSE	1
T39	JETTISON VALVE	46421	LBSEB	BSE	1
T39	JETTISON VALVE	46421	RBSEB	BSE	1
T39	SWITCH, CONTROL	46422	BSEC	BSE	A
T39	VALVE, CROSSFEED	46224	BSED	RSE	0
T39	RELAY, CONTROL	46425	BSEE	RSE	1
T39	RELAY, CONTROL	46425	BSEF	BSE	1
T39	CIRCUIT BREAKER	42152	BSEG	BSE	1
T39	AUTO DUMP		BSF	RSE	111111111
T39	SWITCH, FLOAT	46423	LBSFA	RSF	5
T39	SWITCH, FLOAT	46423	RBSFA	BSF	5

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0000000001111111112222222223333333334444444445555555556666666667777777773
12345678901234567890123456789012345678901234567890123456789012345678901234567890
T39 LIGHT,INDICATOR          46225  BSWA          BSW          A
T39 LIGHT,PANEL              44241  BSWB          BSW          1
T39 LIGHT,MASTER             44242  BSWC          BSW          1
T39 LIGHT,INDICATOR          44243  BSWD          BSW          2
T39 RELAY,TEST                44244  BSWE          BSW          0
T39 CIRCUIT BREAKER          42152  BSWF          BSW          1
T39 FUEL QUANTITY             42152  BSX           B           K BSA  000111100
T39 FUEL QUANTITY             42152  BSX           BSG          AAAAAAAAAA
T39 FUEL QTY INDICATION       42152  BSY           BSX          000111111
T39 INDICATOR,FUEL QTY        51521  LBSYA         BSY          A
T39 INDICATOR,FUEL QTY        51521  RBSYA         BSY          A
T39 TANK UNIT %7 EAC          51522  LBSYB         BSY          1
T39 TANK UNIT %7 EAC          51522  RBSYB         BSY          1
T39 REFERENCE CONDENSER       51523  LBSYC         BSY          A
T39 REFERENCE CONDENSER       51523  RBSYC         BSY          A
T39 THERMISTOR                51525  LBSYD         BSY          A
T39 THERMISTOR                51525  RBSYD         BSY          A
T39 THERMISTOR CONTROL        51526  LBSYE         BSY          A
T39 THERMISTOR CONTROL        51526  RBSYE         BSY          A
T39 TEST SWITCH %SELECTOR<    51524  BSYF          BSY          0
T39 CIRCUIT BREAKER          42231  BSYG          RSY          1
T39 FUEL LEVEL LOW WARNING     42231  BSZ           BSX          K BSY  AAAAAAAAAA
T39 LIGHT,PANEL               44241  BSZA          BSZ          1
T39 LIGHT,MASTER             44242  BSZB          BSZ          1
T39 LIGHT,INDICATOR          44243  BSZC          BSZ          2
T39 FILTER WARNING            44243  BTA           RPF          I RPB  111111111
T39 SW.,DIFF PRESSURE         23218  BTAB          RTA          A
T39 LIGHT,PANEL               44241  BTAC          RTA          1
T39 LIGHT,MASTER             44242  BTAD          BTA          1
T39 LIGHT,INDICATOR          44243  BTAE          BTA          2
T39 CIRCUIT BREAKER          42152  BTAF          BTA          1
T39 COMM/NAV/IDENT            42152  C             C             E       AAAAAAAAAA
T39 COMMUNICATION              42152  CA            C             E       011111120
T39 EXTERNAL COMMUNICATION     42152  CAA           CA            K CAF  111111111
T39 VHF COMMUNICATION          42152  CAC           CAA           K CAF  AAAAAAAAAA
T39 VHF-101                    62100  CACA          CAC           C       0
T39 STATIC DISCHARGER          6211A  CACB          CAC           C       0
T39 ANTENNA                     62111  CACC          CAC           C       8
T39 CONTROL                      62112  CACD          CAC           C       8
T39 MOUNT                        62116  CACE          CAC           C       0
T39 WILCOX 807A                 62200  CACF          CAC           C       0
T39 ANTENNA                     62211  CACG          CAC           C       8
T39 CONTROL                      62112  CACH          CAC           C       8
T39 MOUNT                        62216  CACJ          CAC           C       0
T39 VHF TRANSMIT                62117  CAD           CAD           C       333333333
T39 TRANSMITTER                 62118  CADA          CAD           C       8
T39 TRANSCEIVER                 62217  CADB          CAD           C       5
T39 VHF RECEIVE                  62117  CAE           CAC           C       888888888
T39 RECEIVER                    62117  CAEA          CAF           C       8
T39 TRANSCEIVER                 62217  CAEB          CAF           C       5
    
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T39 UHF COMMUNICATION		CAF	CAA	CAC	111111111	
T39 CONTROL PANEL *C1057B<	6311A	CAFA	CAF		1	
T39 AMPLIFIER OSCILLATOR	6311B	CAFB	CAF		1	
T39 CONTROL MONITOR	6311C	CAFC	CAF		1	
T39 CABINET ELECT EQUIP	6311D	CAFD	CAF		1	
T39 DYNAMOTOR *DY-103<	6311E	CAFE	CAF		1	
T39 MODULATOR RADIO	6311F	CAFF	CAF		1	
T39 SELECTOR CONTROL	6311G	CAFG	CAF		1	
T39 ANTENNA UPPER	6311H	CAFH	CAF		1	
T39 ANTENNA LOWER	6311I	CAFI	CAF		1	
T39 ANTENNA SWITCH	6311J	CAFK	CAF		1	
T39 RELAY ANTENNA CHANGE OVER	6311K	CAFL	CAF		1	
T39 COAXIAL CABLE	6311L	CAFM	CAF		1	
T39 CIRCUIT BREAKER	6311M	CAFN	CAF		1	
T39 POWER SUPPLY *PP-1990<	6311N	CAFO	CAF		1	
T39 MOUNT ASSEMBLY	6311O	CAFQ	CAF		0	
T39 CONTROL PANEL-WIDE BAND	6311P	CAFR	CAF		1	
T39 CABINET ELECT EQUIP	6311Q	CAFS	CAF		1	
T39 MODULATOR RADIO WIDE BAND	6311R	CAFT	CAF		1	
T39 UHF TRANSMIT		CAG	CAF			333333333
T39 RECEIVER TRANS RT463A	6311I	CAGA	CAG		1	
T39 RECEIVER TRANS-WIDE BAND	6311M	CAGB	CAG		1	
T39 UHF RECEIVE		CAH	CAF			888888888
T39 NORMAL RECEIVE		CAJ	CAH			111111111
T39 RECEIVER TRANS-WIDE BAND	6311M	CAJA	CAJ		1	
T39 RECEIVER TRANS RT463A	6311I	CAJB	CAJ		1	
T39 RECEIVER IF-AF R-568	6311H	CAJC	CAJ		1	
T39 RECEIVER IF-AF WIDE BAND	6311Q	CAJD	CAJ		1	
T39 GUARD RECEIVE		CAK	CAH	K CAJ		AAAAAAAAA
T39 RECEIVER GUARD R-567	6311J	CAKA	CAK		8	
T39 SECURE SPEECH		CAL	CAF			000000000
T39 KEYSER DY28	6311R	CALA	CAL		8	
T39 INTERNAL COMMUNICATION		CAM	CA			000000000
T39 STATION TO STATION COMM		CAN	CAA			111111111
T39 STATION TO STATION COMM		CAN	CAM			FAAAAAAAAAA
T39 STATION TO STATION COMM		CAN	CHF			111111111
T39 CIRCUIT BREAKER 3 EACH	6411A	CANA	CAN		1	
T39 HEADSET 3 EACH	6411B	CANB	CAN		1	
T39 U92 PLUG 3 EACH	6411C	CANC	CAN		1	
T39 CORDAGE 3 EACH	6411D	CAND	CAN		1	
T39 CONTROL PANEL 3 EACH	6411I	CANE	CAN		1	
T39 MICROPHONE 3 EACH	64114	CANF	CAN		1	
T39 AUDIO AMPLIFIER 3 EACH	64113	CANG	CAN		1	
T39 RELAY CABIN MIKE 3 EACH	64118	CANH	CAN		1	
T39 SWITCH MICROPHONE	64115	CANJ	CAN		1	
T39 PAX LOUDSPEAKER COMM		CAP	CAM			111111111
T39 CONTROL PANEL 3 EACH	64111	CAPA	CAP		1	
T39 LOUD SPEAKER 2 EACH	64112	CAPB	CAP		1	
T39 SPEAKER AMPLIFIER 2 EACH	64116	CAPC	CAP		1	
T39 EQUIPMENT COOLING		CB	C			000000000

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T39 ELECT EQUIP COOLING SHELF 7111A CBA CB 1
T39 BLOWERS 41410 CBB CB 0
T39 ELECT COMPT GRD BLOWER 41411 CBC CB 0
T39 ELECT SHELF BLOWER 41412 CBD CB 1
T39 HEAT EXCHANGER BLOWER 41416 CBE CB 1
T39 IDENTIFICATION CC C 000000000
T39 AN/APX-72 SYSTEM 65A00 CCA CC 0
T39 SWITCHING UNIT SA-1474 A 65AAA CCB CC 8
T39 MOUNT MT-94817 65AAB CCC CC 0
T39 COMPUTER KIT 1A/TSEC 65AAC CCD CC 1
T39 MOUNT MT-3949A/U 65AAD CCE CC 0
T39 ANTENNA UPPER/LOWER 65AAE CCF CC 1
T39 CO-AXIAL CABLE 65AAF CCG CC 1
T39 CONTROL XPONDER SET 65AAG CCH CC 8
T39 REC-XMITTER RT-859 65AB0 CCJ CC 8
T39 PROCESSOR A1 65ABA CCK CC 8
T39 DECODER A2 65ABB CCL CC 8
T39 MODE 4 A3 65ABC CCM CC 8
T39 ENCODER CLOCK 65ABD CCN CC 8
T39 ENCODER CONTROL 65ABE CCP CC 8
T39 ENCODER GATING 65ABF CCQ CC 8
T39 POWER SUPPLY PSI 65ARG CCR CC 8
T39 MOUNT MT-3509 65ABH CCS CC 0
T39 RF ASSEMBLY 65ABJ CCT CC 8
T39 XPONDER TEST SET TS1843A 65ACO CCU CC 0
T39 MAIN FRAME ASSY A1 65ACA CCV CC 0
T39 COMPARATOR/DECODER A4 65ACB CCW CC 0
T39 MOUNT MT-3513 65ACC CCX CC 0
T39 XPONDER TEST SET TS1843B 65ADO CCY CC 0
T39 SIGNAL GENERATOR/RECEIVER 65ADA CCZ CC 0
T39 DIRECTIONAL COUPLER 65ADB CCZA CC 0
T39 NAVIGATION CD C E 001111140
T39 STEERING SOLUTIONS CE CD 005555580
T39 CIRCUIT BREAKER 71115 CEA CE 1
T39 SWITCH RADIO INST MASTER 71118 CEB CF 1
T39 ENROUTE AIDS CF CF 008888800
T39 ENROUTE NAV DISPLAYS CFA CF 888888888
T39 INDICATOR SWITCHING UNIT 71111 CFAA CFA 5
T39 COURSE SELECTOR SWITCH 71116 CFAB CFA 1
T39 COURSE SEL FAIL LIGHT 71117 CFAC CFA 0
T39 BEARING CFB CFA 888888888
T39 PILOT BEARING CFBA CFB K CFBB 888888888
T39 HORIZ SITUATION IND 51317 CFBA CFBA 8
T39 CIRCUIT BREAKER 42231 CFBA CFBA 1
T39 COPILOT BEARING CFBB CFB K CFBA 888888888
T39 BEARING-DIST-HEAD IND 51314 CFBA CFBB 8
T39 CIRCUIT BREAKER 42152 CFBB CFBB 1
T39 COURSE CFC CFA 111111111
T39 OMNIMAG COURSE IND 51312 CFCA CFC 8
T39 CIRCUIT BREAKER 42231 CFCB CFC 1

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T39 DISTANCE		CFD	000111000
T39 PILOT DISTANCE		CFDA	K CFDB AAAAAAAAA
T39 HORIZ SITUATION IND	51317	CFDAA	8
T39 CIRCUIT BREAKER	42231	CFDAB	1
T39 COPILOT DISTANCE		CFDB	K CFDA AAAAAAAAA
T39 BEARING-DIST-HEAD IND	51314	CFDBA	8
T39 CIRCUIT BREAKER	42152	CFDBB	1
T39 HEADING		CFE	444444444
T39 PILOT HEADING		CFEA	K CFEB AAAAAAAAA
T39 ATTITUDE DIRECT IND	51232	CFEAA	8
T39 CIRCUIT BREAKER	42231	CFEAB	1
T39 COPILOT HEADING		CFEB	K CFEA AAAAAAAAA
T39 COPILOT 1		CFEC	111111111
T39 BEARING-DIST-HEAD IND	51314	CFECA	8
T39 CIRCUIT BREAKER	42152	CFECB	1
T39 COPILOT 2		CFED	111111111
T39 OMNIMAG COURSE IND	51312	CFEDA	8
T39 CIRCUIT BREAKER	42231	CFEDB	1
T39 ENROUTE NAV ELECTRONICS		CFE	AAAAAAAAA
T39 TACAN		CFG	111111111
T39 MODULATOR MD-359	7121A	CFG	8
T39 AZIMUTH GATE TD-273	7121B	CFG	8
T39 AMPLIFIER AM-2212	7121D	CFG	8
T39 RT UNIT RF-98	7121E	CFG	8
T39 POWER SUPPLY PP-2331	7121F	CFG	8
T39 VIDEO DECODER KY-290	7121G	CFG	8
T39 PRESELECTOR Z108	7121K	CFG	0
T39 CHANNEL SELECTOR C-2875	7121L	CFG	8
T39 CRYSTAL SELECTOR TG-68	7121M	CFG	A
T39 CIRCUIT BREAKER	42231	CFG	1
T39 REC/TRANSMIT		CFH	AAAAAAAAA
T39 ANTENNA	7121N	CFHA	A
T39 REC/XMITTER RT-220C	71211	CFHB	8
T39 CONTROL PANEL C-1763	71212	CFHC	8
T39 MOUNT ASSY MT-929	71216	CFHD	0
T39 SHOCK ASSY	71217	CFHE	0
T39 RADIO INST MASTER RELAY3/471114		CFHF	1
T39 RANGE		CFJ	000111000
T39 RANGE CONTROL	7121C	CFJA	8
T39 RANGE GATE TD-272	7121H	CFJB	8
T39 COUPLING		CFK	888888888
T39 TACAN COUPLER	71220	CFKA	8
T39 IND COUPLER	71221	CFKB	8
T39 RANGE MODULE	71222	CFKC	1
T39 AZIMUTH MODULE	71223	CFKD	8
T39 PHASE DETECTOR	71224	CFKE	8
T39 MOUNTING	71225	CFKF	0
T39 VOR		CFL	111111111
T39 VOR		CFL	AAAAAAAAA
T39 CONTROL PANEL	71512	CFLA	8

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T39	INST MASTER RELAYS 3 AND	571114	CFLB CFL 1
T39	CIRCUIT BREAKER	42152	CFLC CFL 1
T39	INSTRUMENTATION		CFM CFL AAAAAAAAAA
T39	INSTRUMENTATION UNIT	71514	CFMA CFM 8
T39	POWER SUPPLY AC 516A-1	71515	CFMB CFM 8
T39	POWER SUPPLY AC 516B-3	71516	CFMC CFM 8
T39	RF-1F AMP MODULE 51X2B	71517	CFMD CFM 8
T39	1F AND AUDIO MODULE	71518	CFME CFM 8
T39	503KC FILTER MODULE	7151A	CFMF CFM 3
T39	CIRCUIT BREAKER	42231	CFMG CFM 1
T39	RECEIVE		CFN CFL AAAAAAAAAA
T39	REF AMP MODULE 344B1	71518	CFNA CFN 8
T39	VARIABLE AMP MODULE	7151C	CFNB CFN 8
T39	CONVERT DISCRIM MODULE	7151E	CFNC CFN 8
T39	BEARING SERVO MODULE	7151F	CFND CFN 8
T39	VDR FLAG CONT MODULE	7151H	CFNE CFN 1
T39	BEARING MECHANISM MODULE	7151J	CFNF CFN 8
T39	AC POWER SUPPLY MODULE	7151K	CFNG CFN 8
T39	CHASSIS ASSY MODULE	7151L	CFNH CFN 1
T39	MOUNT ASSY	7151M	CFNJ CFN 0
T39	RECEIVER	71511	CFNK CFN 8
T39	ANTENNA	71513	CFNL CFN A
T39	REMOTE DIRECTIONAL GYRO		CG CE 888888888
T39	RELAY	7171A	CGA CG 1
T39	REPEATER AMP A5	7171B	CGB CG 3
T39	REPEATER MECHANISM MOD A-6	7171C	CGC CG 8
T39	DIRECTIONAL GYRO	71712	CGD CG A
T39	COMPASS AMPLIFIER	71714	CGE CG 8
T39	SHOCK MOUNT	71715	CGF CG 0
T39	SERVO AMP A1	71716	CGG CG 8
T39	PRE AMP A2	71717	CGH CG 8
T39	POWER SUPPLY A3	71718	CGJ CG 8
T39	GYRO	71912	CGK CG A
T39	COUPLER	71914	CGL CG 8
T39	MOUNT	71915	CGM CG 0
T39	SLAVING AMP	71916	CGN CG 8
T39	POWER SUPPLY	71917	CGP CG 8
T39	MECHANISM COMPASS	71918	CGQ CG 8
T39	REMOTE MAG FLUX DETECT	51315	CGR CG 3
T39	GYRO MODE SELECT SWITCH	51322	CGS CG 5
T39	APPROACH AND LANDING AIDS		CH CE 000000000
T39	VORZILS		CHA CH 111111111
T39	HEADING		CHB CHA AAAAAAAAAA
T39	PILOT HEADING		CHBA K CHBB 111111111
T39	ATTITUDE DIRECTOR IND	51232	CHBAA CHBA 8
T39	CIRCUIT BREAKER	42231	CHBAB CHBA 1
T39	COPILOT HEADING		CHBB K CHBA AAAAAAAAAA
T39	COPILOT 1		CHBC CHBB 111111111
T39	BEARING-DIST-HEAD IND	51314	CHBCA CHBC 8
T39	CIRCUIT BREAKER	42152	CHBCB CHBC 1

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T39 COPILOT 2		CHBD	CHBR		111111111
T39 OMNIMAG COURSE IND	51312	CHBDA	CHBD	8	
T39 CIRCUIT BREAKER	42231	CHBDB	CHBD	1	
T39 GLIDESLOPE		CHC	CHA		AAAAAAAAA
T39 POSER SUPPLY 51V-3	7141A	CHCA	CHC	8	
T39 RF OSCILLATOR	7141E	CHCB	CHC	8	
T39 POSER SUPPLY DGS-20	7141F	CHCC	CHC	8	
T39 TONE FILTER	7141H	CHCD	CHC	1	
T39 RECEIVER	71411	CHCE	CHC	8	
T39 ANTENNA	71412	CHCF	CHC	A	
T39 CONTROL PANEL	71413	CHCG	CHC	8	
T39 MOUNT ASSY	71414	CHCH	CHC	0	
T39 PILOT GLIDESLOPE		CHCX	CHC	K CHCY	AAAAAAAAA
T39 ATTITUDE DIRECTOR IND	51232	CHCXA	CHCX	8	
T39 CIRCUIT BREAKER	42231	CHCXB	CHCX	1	
T39 COPILOT GLIDESLOPE		CHCY	CHC	K CHCX	AAAAAAAAA
T39 OMNIMAG COURSE IND	51312	CHCYA	CHCY	8	
T39 CIRCUIT BREAKER	42231	CHCYB	CHCY	1	
T39 VOR RECEIVE		CHD	CHA		AAAAAAAAA
T39 LOCALIZER OUTPUT MODULE	7151G	CHDA	CHD	8	
T39 ADF		CHE	CH	K CHA	AAAAAAAAA
T39 ADF RECEIVE		CHEA	CHE		AAAAAAAAA
T39 ANTENNA	71811	CHEAA	CHEA	A	
T39 CONTROL	71812	CHEAB	CHEA	8	
T39 INDICATOR	71813	CHEAC	CHEA	1	
T39 DYNAMOTOR	71814	CHEAD	CHEA	A	
T39 RECEIVER	71815	CHEAE	CHEA	8	
T39 MOUNT	71818	CHEAF	CHEA	0	
T39 CIRCUIT BREAKER	42152	CHEAG	CHEA	1	
T39 CIRCUIT BREAKER	42231	CHEAH	CHEA	1	
T39 BEARING		CHER	CHE		AAAAAAAAA
T39 PILOT BEARING		CHEC	CHEB	K CHED	AAAAAAAAA
T39 HORIZ SITUATION IND	51317	CHECA	CHEC	8	
T39 CIRCUIT BREAKER	42231	CHECB	CHEC	1	
T39 COPILOT BEARING		CHED	CHEB	K CHEC	AAAAAAAAA
T39 BEARING-DIST-HEAD IND	51314	CHEDA	CHED	8	
T39 CIRCUIT BREAKER	42152	CHEDB	CHED	1	
T39 MARKER BEACON		CHF	CH		000000000
T39 RJ AND IF MODULE 5122	7161A	CHFA	CHF	8	
T39 AMPLIFIER MODULE 5122	7161B	CHFB	CHF	8	
T39 RECEIVER	71611	CHFC	CHF	8	
T39 ANTENNA	71612	CHFD	CHF	A	
T39 SENSITIVITY SWITCH	71613	CHFE	CHF	1	
T39 INDICATOR LIGHT	71614	CHFF	CHF	1	
T39 SWITCH TEST	71615	CHFG	CHF	0	
T39 DIM AND TEST RELAY BOX	71112	CHFH	CHF	0	
T39 FLIGHT DIRECTOR GROUP		CJ	CF		111111111
T39 COMPUTER	71311	CJA	CJ	8	
T39 FLT DIRECTOR SELECT SW	71312	CJB	CJ	5	
T39 MOUNT ASSY	71313	CJC	CJ	0	

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T39 DEAD RECONING                CK                CD      K CE      000000000
T39 HEADING                       CKA             CK                000000000
T39 STANDBY COMPASS                51311          CKA             CKA             1
T39 COMPASS CORRECTION CARD        51324          CKAB            CKA             0
T39 SEXTANT                        51316          CKAC            CKA             1
T39 SEXTANT MOUNT                  51318          CKAD            CKA             0
T39 TIME                           CKB            CK                000000000
T39 CLOCK                          51321          CKBA            CKB             0
T39 WEATHER AVOIDANCE              CL              CD              000000000
T39 MODULATOR                      7221A          CLA             CL              8
T39 REC-XMITTER RT-1018-28         72221          CLB             CL              8
T39 INDICATOR IN-112               72212          CLC             CL              8
T39 ANTENNA AT-113                 72213          CLD             CL              A
T39 MOUNT MT-104AH                 72214          CLE             CL              0
T39 INFO AND DISPLAY               D              D              000000000
T39 FLIGHT STATUS                  DA             D              011111130
T39 ATTITUDE                       DAA            DA              E              000000000
T39 CIRCUIT BREAKER                42231          DAAA           DAA             1
T39 INST PANEL VIBRATOR 2 EA       51111          DAAAA          DA              0
T39 SHOCK MOUNT                    51112          DAAAAB         DA              0
T39 CKT BKR VIBRATOR               51113          DAAAC          DA              0
T39 TURN-SLIP INDICATION           DAB            DAA            K DAC          000000000
T39 INSTR PANEL COPILOT OUTBRD     12112          DABA           DAB             1
T39 RATE GYRO TRANSMITTER          5123A          DABB           DAB             8
T39 TURN AND SLIP INDICATOR        51233          DABC           DAB             A
T39 CIRCUIT BREAKER                42152          DABD           DAB             1
T39 ROLL-PITCH                     DAC            DAA            DAA            999999999
T39 PILOT INDICATION              DAD            DAC            K DAE          000000000
T39 INSTR PANEL PILOT OUTBRD       12111          DADA           DAD             1
T39 ATTITUDE DIR INDICATOR         51232          DADP           DAD             A
T39 VERTICAL GYRO MD1               51234          DADC           DAD             A
T39 RATE SWITCHING GYRO MC1        51235          DADD           DAD             A
T39 COPILOT INDICATION             DAE            DAC            K DAD          000000000
T39 INSTR PANEL COPILOT OUTBRD     12115          DAEA           DAE             1
T39 ATTITUDE INDICATOR             51231          DAEB           DAE             A
T39 VERTICAL GYRO MD1               51234          DAEC           DAE             A
T39 RATE SWITCHING GYRO MC1        51235          DAED           DAE             A
T39 ALTITUDE INFO                  DAF            DA              E              0001111AA0
T39 VERTICAL VELOCITY              DAG            DAF            DAF            000000010
T39 PILOT INDICATION              DAH            DAG            K DAJ          000000000
T39 INSTR PANEL PILOT OUTBRD       12111          DAHA           DAH             1
T39 VERT VELOCITY INDICATOR        51224          DAHB           DAH             A
T39 COPILOT INDICATION             DAJ            DAG            K DAH          000000000
T39 INSTR PANEL COPLT OUTBRD       12115          DAJA           DAJ             1
T39 VERT VELOCITY INDICATOR        51224          DAJB           DAJ             A
T39 ALTITUDE INDICATION            DAK            DAF            DAF            000000000
T39 COMMUTER CPU-46A B26           51238          DAKA           DAK             0
T39 MOUNT                           5123C          DAKB           DAK             0
T39 COMPUTER CPU 46A/A             5123D          DAKC           DAK             0
T39 PILOT INDICATION              DAL            DAK            K DAM          000000000
    
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FLIGHT SAFETY PREDICTION TECHNIQUE

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0000000001111111111222222222333333333444444444555555555666666667777777778
1234567890123456789012345678901234567890123456789012345678901234567890123456789012345678901234567890
T39 INSTR PANEL PILOT OUTBRD 12111 DALA DAL 1
T39 ALTIMETER 51223 DALB DAL A
T39 ALTIMETER AAU 19A 51227 DALC DAL A
T39 ALTIMETER AAU 19AA 51228 DALD DAL A
T39 COPILOT INDICATION DAM DAK K DAL AAAAAAAAAA
T39 INST PANEL COPLT OUTBRD 12115 DAMA DAM 1
T39 ALTIMETER 51223 DAMB DAM A
T39 ALTIMETER AAU 19A 51227 DAMC DAM A
T39 ALTIMETER AAU 19AA 51228 DAMD DAM A
T39 NAVIGATOR INDICATION DAN DAK 000000000
T39 ALTIMETER 51223 DANA DAN A
T39 ALTIMETER 51227 DANB DAN A
T39 ALTIMETER 51228 DANC DAN A
T39 ATTENUATION DAP DA 111111111
T39 G-LOAD DAQ DAP 000000000
T39 INSTR PANEL COPLT OUTBRD 12115 DAQA DAQ 1
T39 ACCELEROMETER 51211 DAQB DAQ A
T39 AIRSPEED DAR DAP GA1000140
T39 SYSTEM NO 1 INDICATION DAS DAK K DAW AAAAAAAAAA
T39 INSTR PANEL PILOT OUTBRD 12111 DASA DAS 1
T39 PILOT INDICATED AIR SPEED DAT DAS 111111111
T39 AIRSPEED INICATOR 51221 DATA DAT A
T39 PILOT MACHMETER DAU DAS 111111111
T39 MACHMETER 51222 DAUA DAU A
T39 MACH-AIRSPEED WARNING DAV DAS 000000000
T39 CIRCUIT BREAKER 42152 DAVA DAV 1
T39 HORN MACH AIRSPEED 51212 DAVB DAV A
T39 CONTROL MACH AUDIBLE SIG 51213 DAVC DAV 1
T39 SWITCH AIRSPEED MACH 51248 DAVD DAV A
T39 SYSTEM NO 2 INDICATION DAW DAR K DAS AAAAAAAAAA
T39 COPILOT INDICATED AIRSPEED DAX DAW 555555555
T39 INSTR PANEL COPLT OUTBRD 12115 DAXA DAX 1
T39 AIRSPEED INDICATOR 51221 DAXB DAX A
T39 NAVIGATOR TRUE AIRSPEED DAY DAW K DAX AAAAAAAAAA
T39 AIRSPEED INDICATOR 51221 DAYA DAY A
T39 CORRECTED AIRSPEED DAZ DAY AAAAAAAAAA
T39 STATIC NO 1 DIST DBA DAH AAAAAAAAAA
T39 STATIC NO 1 DIST DBA DAL AAAAAAAAAA
T39 STATIC NO 1 DIST DBA DAS AAAAAAAAAA
T39 STATIC NO 1 PRESSURE DBB DBA DBE 111111111
T39 DRAIN PLUG #2 EAC 51245 DB6A DBB 0
T39 LH PITOT-STATIC #STATIC< DBC DBB 111111111
T39 STATIC PORT 51244 DBCA DBC A
T39 RH PITOT-STATIC #STATIC< DBD DBB 111111111
T39 STATIC PORT 51244 DBDA DBD A
T39 ALTERNATE SELECT NO 1 DBE DBA K DBB AAAAAAAAAA
T39 STATIC SELECTOR VALVE 51243 DBEA DBE A
T39 ALTERNATE STATIC DBF DBE AAAAAAAAAA
T39 ALTERNATE STATIC DBF DBH AAAAAAAAAA
T39 ALTERNATE STATIC DBF NAT 000000000
  
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00000000011111111122222222333333333344444444445555555555666666666677777777778
12345678901234567890123456789012345678901234567890123456789012345678901234567890
T39 CABIN LIGHTING EAB EAA 00000C000
T39 CONTROL PANEL 44211 EABA FAB 1
T39 SWITCH 10 EACH 44212 EABB FAB 1
T39 LIGHT ASSY 8EA 44215 EABC EAB 1
T39 LIGHT ASSY SIGN 2EA 44216 EABD EAB 1
T39 COCKPIT LIGHTING EAC EAA AAAAAAAAAA
T39 EMERGENCY LIGHTING EAD EAC K EAF AAAAAAAAAA
T39 RELAY CONTROL 44218 EADA EAD A
T39 POWER UNIT-EMERGENCY 4421E EADB EAD A
T39 LIGHT ASSY 3EA 44215 EADC EAD 1
T39 NORMAL LIGHTING EAE FAC EAD 111111111
T39 CIRCUIT BREAKER DC 42152 EAEA EAF 1
T39 CIRCUIT BREAKER AC 42231 EAEB EAF 1
T39 INSTRUMENT LIGHTS EA OF 2 EAF EAF 111111111
T39 RESISTOR DIMMING 4421C EAFA EAF 1
T39 CONTROL PANEL 44211 EAFB EAF 1
T39 LIGHT ASSY FLOOD 44215 EAFD EAF 1
T39 LIGHT ASSY INSTR 44231 EAFD EAF 1
T39 CONSOLE LIGHTS EA OF 2 EAG EAE 000000000
T39 RESISTOR DIMMING 4421C EAGA EAG 1
T39 CONTROL PANEL 44211 EAGR EAG 1
T39 LIGHT ASSY 44215 EAGC EAG 1
T39 DOME LIGHTS EAH EAF 000000000
T39 RESISTOR DIMMING 2EA 4421C EAHA EAH 1
T39 CONTROL PANEL 44211 EAHB EAH 1
T39 SWITCH 2EA 44212 EAHC FAH 1
T39 LIGHT ASSY 44215 EAHD EAH 1
T39 EXTERNAL ATTENUATION EAL EA 111111121
T39 EXTERNAL LIGHTING EAM EAL 111111111
T39 CIRCUIT BREAKER 42152 EAMA EAM 1
T39 LAND/TAXI LIGHTS EAN EAM 100000051
T39 CONTROL PANEL 44112 EANA EAN 1
T39 LANDING AND TAXI LT ASSY 44113 EANB EAN 5
T39 LAND AND TAXI LT BRACKET 44117 EANC EAN 1
T39 SWITCH 44118 EAND EAN 5
T39 POSITION LIGHTS EAP EAM 000000000
T39 CONTROL PANEL 44112 EAPA EAP 1
T39 POSITION LIGHT ASSY 6EA 44114 EAPB EAP 1
T39 SWITCH 44118 EAPC EAP 5
T39 ANTI-COLLISION LIGHT EAQ EAM 000000000
T39 CONTROL PANEL 44112 EAQA EAQ 1
T39 ANTI-COLLISION LT ASSY 2EA 44115 EAQB EAQ 1
T39 SWITCH 44118 EAQC EAQ 5
T39 ICE CHECK LIGHTS EAR EAM 000000000
T39 CONTROL PANEL 44112 EARA EAR 1
T39 ICE CHECK LIGHT ASSY 2EA 44116 EARB EAR 1
T39 SWITCH 44118 EARC EAR 5
T39 CIRCUIT BREAKER 42152 EARD EAR 1
T39 WINDSHIELD CLEARING EB EBC GA00000GA0
T39 RAIN REMOVAL ERA EB G 010000010

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T39	BLADE 2EA	49411	EBAA	ERA	1
T39	ARM ASSY 2EA	49412	EBAB	ERA	1
T39	CONVERTER ASSY	49413	EBAC	ERA	8
T39	FLEX DRIVE	49414	EBAD	FBA	A
T39	MOTOR ASSY	49415	EBAE	EBA	8
T39	CIRCUIT BREAKER	42152	EBAF	EBA	1
T39	ANTI-ICE		EBB	EB	A 010000010
T39	WINDSHIELD HEAT	41530	EBBA	EBB	0
T39	HEATING ELEMENT 6EA	41531	EBBR	EBB	1
T39	SENSING ELEMENT 5EA	41532	EBBC	EBB	1
T39	OVERHEAT THERMOSWITCH	41533	EBBD	EBB	5
T39	CONTROLLER	41535	EBBE	EBB	3
T39	SWITCH	41536	EBBF	FRB	A
T39	AC GENERATOR SYSTEM	42210	EBBG	EBB	0
T39	GENERATOR ENG DRIVEN 2EA	42211	EBBH	EBB	1
T39	ADAPTER QAD	42212	EBBJ	EBB	8
T39	LIGHT PANEL	44241	EBBK	EBB	1
T39	LIGHT MASTER	44242	EBBL	FRB	1
T39	LIGHT INDICATOR	44243	EBBM	EBB	1
T39	CIRCUIT BREAKER	42152	EBBN	EBB	1
T39	ATTENUATION		EBC	E	010000020
T39	CABIN ENVIRONMENT		EC	F	111111111
T39	AIR COND AND PRESSURIZATION		ECA	EC	ED 001111100
T39	COCKPIT AIR OUTLET 4EA	41116	ECAA	ECA	0
T39	CABIN AIR OUTLET 12EA	41117	ECAB	ECA	0
T39	CAB GRD AIR COND FIT-CK V	41215	ECAC	ECA	0
T39	CIRCUIT BREAKER 7EA	42152	ECAD	ECA	1
T39	CABIN AIR DUMP MAN CONT VL	41213	ECAE	ECA	0
T39	HOT AIR SUPPLY		ECB	ECA	AAAAAAAAA
T39	EMERGENCY DUCTING		ECC	ECH	K ECD AAAAAAAAAA
T39	EMER PRESS BLED SHTOFF VAL	4111E	ECCA	ECC	A
T39	PRESS DUCTFAIL SWITCH	9941B	ECCB	ECC	1
T39	DUCT FAIL LIGHT	44242	ECCC	FCC	1
T39	NORMAL HOT AIR		ECD	ECH	ECC 111111111
T39	BLEED AIR FLO LIMIT VALV	4111C	ECDA	FCD	1
T39	DUCTING	41114	ECDB	ECD	1
T39	SEALING		ECE	ECA	111111111
T39	REV FLO CK VALV 2EA	4111F	ECEA	ECE	1
T39	DOOR SEAL PRESS RESERVOIR	41311	ECER	ECE	1
T39	PNEUMATIC PRESS CK VALV	41312	ECEC	ECF	1
T39	ESCAPE HATCH SEAL	41313	ECED	ECE	3
T39	ESCAPE HATCH SEAL NIPPLE	41315	ECEE	ECE	0
T39	ENTRANCE DOOR SEAL	41316	ECEF	ECE	3
T39	REGULATOR ASSY DOOR SEAL	41317	ECEG	ECE	8
T39	RAIN SEAL ENTRANCE DOOR	11215	ECEN	ECE	1
T39	PRESSURE REGULATION		ECF	ECA	333333333
T39	AUTO REGULATION		ECG	ECF	ECH 111111111
T39	CAB AIR PRESS REG	41211	ECGA	ECG	A
T39	PRESSURE DUMP		ECH	ECF	K ECG AAAAAAAAAA
T39	CAB AIR DUMP-SAFETY VALV	41212	ECHA	ECH	1

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1234567890123456789012345678901234567890123456789012345678901234567890
T39 LIGHT INDICATOR                    44243      ECUD                    ECU                    1
T39 RIGHT ENGINE BLEED AIR            ECV                    ECC                    AAAAAAAAAA
T39 RIGHT ENGINE BLEED AIR            ECV                    ECD                    K ECW                AAAAAAAAAA
T39 ENG BLEED SHUTOFF VALVE          4111D      ECVA                    ECV                    5
T39 REVERSE FLO CK VALV              4111F      ECVB                    ECV                    1
T39 DUCTING                            41114      ECVC                    ECV                    1
T39 LEFT ENGINE BLEED AIR            ECV                    ECD                    K ECV                AAAAAAAAAA
T39 ENGINE BLEED SHUTOFF VALVE 4111D      ECWA                    ECW                    5
T39 REV FLOW CHECK VALVE            4111F      ECWB                    ECW                    1
T39 DUCTING                            41114      ECWC                    ECW                    1
T39 DESCENT SUPP PRESSURE            ECX                    ECF                    000000100
T39 CAB FLOOD FLOW VALVE            41127      ECXA                    ECX                    A
T39 PRESSURE INTEGRITY                ECV                    ECE                    111111111
T39 PILOTS SLIDING WINDOW ASSY 11120      ECYA                    ECV                    8
T39 COCKPIT WINDOW ASSY              1114C      ECYB                    ECV                    0
T39 CABIN WINDOW ASSY                11170      ECYC                    ECV                    0
T39 ENTRANCE DOOR ASSY               11210      ECYD                    ECV                    8
T39 EMER ESCAPE INNER DR ASSY 11310      ECYE                    ECV                    3
T39 EMER ESCAPE OUTER DR ASSY 11320      ECYF                    ECV                    3
T39 EMER ESCAPE JETTISON SYST 11330      ECYG                    ECV                    0
T39 HACKUP                            ED                    K ECA                00AAAAAAAA0
T39 VENTILATION                        EDA                    ED                    333333333
T39 EMERG RAM AIR SUPPLY VALV 41222      EDAA                    EDA                    A
T39 CIRCUIT BREAKER                  42152      EDAB                    FDA                    1
T39 RAM AIR INLET VALVE              41520      EDAC                    EDA                    A
T39 HEATING ELEMENT                 41521      EDAD                    EDA                    0
T39 BIRD STOPPER FLAPPER DOOR 41522      EDAE                    EDA                    1
T39 OXYGEN                            EE                    ED                    AAAAAAAAAA
T39 BACKUP OXYGEN                    EEA                    EE                    K EEB                AAAAAAAAAA
T39 RECHARGER HOSE ASSY              4711M      EEAA                    EEA                    0
T39 PORTABLE OXYGEN CYLINDER        47211      EEAB                    EEA                    A
T39 GAGE                               47212      EEAC                    EEA                    0
T39 SUPPORT BRACKET                  47213      EEAD                    EEA                    0
T39 REGULATOR                        47214      EEAE                    EEA                    8
T39 FIXED OXYGEN                      EEB                    EE                    EEA                222222222
T39 SUPPLY                            EEC                    EEB                    EEB                    AAAAAAAAAA
T39 MANIFOLD                          4711P      EECA                    EEC                    3
T39 OXYGEN CYLINDER                  47112      EECB                    EEC                    A
T39 FILLER VALVE                      47113      EECC                    EEC                    1
T39 PRESSURE REDUCER                47115      EECD                    EEC                    1
T39 COCKPIT OXYGEN                    EED                    EEB                    EEB                    AAAAAAAAAA
T39 REGULATOR PILOTS 2EA            4711D      EEDA                    EED                    1
T39 MASK 2EA                          4711F      EEDB                    EED                    1
T39 AIR BREATHING VALVE 2EA         4711G      EEDC                    EED                    1
T39 MASK-TO-REG HOSE 2EA            4711J      EEDD                    EED                    1
T39 CABIN OXYGEN                      EEE                    EEB                    EEB                    000000000
T39 REGULATOR PASSENGER 4EA        4711E      EEEA                    EEE                    1
T39 MASK 7EA                          4711F      EEEB                    EEE                    1
T39 AIR BREATHING VALVE 4EA         4711G      EEFC                    EEE                    1
T39 IN-USE-VALVE 7EA                 4711H      EEED                    EEE                    1

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T39 MASK-TO-RFG HOSE 7EA	4711J	EEEE	EEF	1
T39 COMPARTMENT DOORS 7EA	4711K	EEEF	EEF	0
T39 OXYGEN SHUTOFF VALVE	4711A	EEEG	EEF	A
T39 OXYGEN CONTROL VALVE	4711B	EEEH	EEF	A
T39 INDICATION		EEG	EEA	111111111
T39 CIRCUIT BREAKER	42152	EEGA	EEG	1
T39 CIRCUIT BREAKER	42231	EEGB	EFG	1
T39 PRESSURE SWITCH	4711B	EEGC	EEG	A
T39 INDICATOR LIGHT	4711C	EEGD	EEG	1
T39 FLOW INDICATOR 2EA	4711N	EEGE	EEG	1
T39 PRESSURE TRANSMITTER	47116	EEGF	EEG	1
T39 PRESSURE GAGE	47117	EEGG	EEG	1
T39 CABIN FIRE EXTINGUISH		EF	E	X AAAAAAAAAA
T39 PORTABLE SYSTEM	49230	EFA	EF	8
T39 BRACKET ASSY	49231	EFB	EF	0
T39 GAGE	49232	FFC	EF	0
T39 EXTINGUISHER	49233	FFD	EF	8
T39 FLIGHT CONTROL		F		AAAAAAAAAA
T39 LIFT AUGMENTATION		FA	F	01000030
T39 FLAPS POSITIONING		FAA	FA	AAAAAAAAAA
T39 INDICATOR, POSITION	51611	FAAA	FAA	0
T39 XMITTER, POSITION	51612	FAAB	FAA	0
T39 CIRCUIT BREAKER	42152	FAAC	FAA	1
T39 LEFT FLAP POSITIONING		LFAB	FAA	AAAAAAAAAA
T39 RIGHT FLAP POSITIONING		RFAB	FAA	AAAAAAAAAA
T39 WING FLAP ASSY	14610	LFABA	LFAB	1
T39 WING FLAP ASSY	14610	RFABA	RFAB	1
T39 SKIN	14612	LFABB	LFAB	0
T39 SKIN	14612	RFABB	RFAB	0
T39 HONEYCOMB	14613	LFABC	LFAB	0
T39 HONEYCOMB	14613	RFABC	RFAB	0
T39 FRAME STRUCTURE	14615	LFABD	LFAB	1
T39 FRAME STRUCTURE	14615	RFABD	RFAB	1
T39 INBOARD ROLLER ASSY	14616	LFABE	LFAB	3
T39 INBOARD ROLLER ASSY	14616	RFABE	RFAB	3
T39 OUTBOARD ROLLER ASSY	14617	LFABF	LFAB	3
T39 OUTBOARD ROLLER ASSY	14617	RFABF	RFAB	3
T39 TRACK ASSY	14631	LFABG	LFAB	3
T39 TRACK ASSY	14631	RFABG	RFAB	3
T39 LEFT FLAP ACTUATION		LFAC	FAA	KRFAC 00AAAAAAAA
T39 RIGHT FLAP ACTUATION		RFAC	FAA	KLFAC 00AAAAAAAA
T39 ACTUATOR	14635	LFACA	LFAC	A
T39 ACTUATOR	14635	RFACA	RFAC	A
T39 FLEX SHAFT	14632	FACB	LFAC	A
T39 FLEX SHAFT	14632	FACB	RFAC	A
T39 INTERCONNECT	14634	FACC	LFAC	A
T39 INTERCONNECT	14634	FACC	RFAC	A
T39 FLAPS CONTROL		FAD	FAA	00AAAAAAAA
T39 SW., FLAPS	9914A	FADA	FAD	A
T39 SLATS POSITIONING ATTENUATE		FAE	FA	111111111

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1234567890123456789012345678901234567890123456789012345678901234567890
T39 PITCH CONTROL FC F OAAAAAAAAA
T39 ELEVATOR POSITIONING FCA FC AAAAAAAAAA
T39 ELEVATOR CONTROL FCB FCA 094444490
T39 PILOT CONT WHEEL/COLUMN 14111 FCBA FCB 1
T39 COPILOT CONT WHEEL/COLUMN 14112 FCBB FCB 1
T39 BOB WEIGHT BUNGEE 14436 FCBC FCB 0
T39 TORQUE TUBE 14437 FCBD FCB 8
T39 LEFT ELEVATOR DRIVE LFCC FCB KR FCC AAAAAAAAAA
T39 RIGHT ELEVATOR DRIVE RFCC FCB KL FCC AAAAAAAAAA
T39 CABLE 14434 LFCCA LFCC A
T39 CABLE 14434 RFCCA RFCC A
T39 LINKAGE 14433 LFCCB LFCC 8
T39 LINKAGE 14433 RFCCB RFCC 8
T39 HORN 14438 LFCCC LFCC 1
T39 HORN 14438 RFCCC RFCC 1
T39 LEFT ELEVATOR POSITIONING LFCD FCB AAAAAAAAAA
T39 RIGHT ELEVATOR POSITIONING RFCD FCB AAAAAAAAAA
T39 ELEVATOR ASSY 14410 LFCDA LFCD 1
T39 ELEVATOR ASSY 14410 RFCDA RFCD 1
T39 SKIN 14412 LFCDB LFCD 0
T39 SKIN 14412 RFCDB RFCD 0
T39 HINGE SUPPORT 14415 LFCDC LFCD 2
T39 HINGE SUPPORT 14415 RFCDC RFCD 2
T39 HOR STAB TRIM POSITIONING FCE FCA 011000110
T39 TORQUE TUBE 14326 FCEA FCE 8
T39 STAR POS INDICATOR 51625 FCEB FCE 0
T39 STAR POS XMITTER 51626 FCEC FCE 0
T39 CENTER SECTION BEAM 14325 FCED FCE 8
T39 BIAS BUNGEE 14435 FCEE FCE 1
T39 FLEX SHAFT ASSY 14324 FCEF FCF A
T39 CIRCUIT BREAKER 42152 FCEG FCE 1
T39 LEFT TRIM ACTUATION LF CF FCE KR FCF AAAAAAAAAA
T39 RIGHT TRIM ACTUATION RF CF FCE KL FCF AAAAAAAAAA
T39 ACTUATOR TRIM 14322 LFCFA LF CF A
T39 ACTUATOR TRIM 14322 RFCFA RF CF A
T39 STABILIZER POSITIONING FA 2 FCG FCE AAAAAAAAAA
T39 HOR STAB ASSY 14310 FCGA FCG 1
T39 SKIN 14312 FCGB FCG 0
T39 NORMAL TRIM ACTIVATION FCH FCE FCJ 111111111
T39 SW., TRIM SELECTOR 14117 FCHA FCH A
T39 SW., TRIM %NORMAL < %2EA< 14113 FCHB FCH 1
T39 SW., EMER DISCONNECT %2EA< 14116 FCHC FCH 1
T39 ALTERNATE TRIM FCJ FCE K FCH AAAAAAAAAA
T39 SW., TRIM SELECTOR 14117 FCJA FCJ A
T39 SW., TRIM %ALTERNATE < 14115 FCJB FCJ A
T39 ROLL CONTROL FD F JAAAAAAAAA
T39 AILERON POSITIONING FDA FD AAAAAAAAAA
T39 AILERONS POSITIONED, FA OF 2 FDB FDA 021111120
T39 AILERON ASSY 14210 FD BA 1
T39 SKIN 14212 FD BB 0
    
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T39	HONEYCOMB	14213	FD8C	FDB	0				
T39	CABLE	14235	FD8D	FDB	A				
T39	PUSH ROD	14232	FD8E	FDB	A				
T39	BELLCRANK	14231	FD8F	FDB	7				
T39	AILERON CONTROL		FDC	FDA				AAAAAAAA	
T39	BELLCRANK	14231	FDCA	FDC	A				
T39	LINKAGE	14234	FDCB	FDC	A				
T39	PILOT CONTROL		FDD	FDC		K FDE		AAAAAAAA	
T39	CONTROL WHEEL/COLUMN	14111	FDDA	FDD	A				
T39	COPILOT CONTROL		FDE	FDC		K FDD		AAAAAAAA	
T39	CONTROL WHEEL/COLUMN	14112	FDEA	FDE	A				
T39	TRIM		FDF	FDA				00000000	
T39	TRIM TAB	14214	FDF A	FDF	A				
T39	TAB ACCESS DOOR	14218	FDFB	FDF	0				
T39	TRIM ACTUATOR	14236	FDFC	FDF	A				
T39	AILRON TRIM INDICATOR	51621	FDFD	FDF	0				
T39	AILRON POSITION XMITTER	51622	FDFE	FDF	0				
T39	SW., TRIM SELECTOR	14117	FDF F	FDF	A				
T39	CIRCUIT BREAKER	42152	FDFG	FDF	1				
T39	NORMAL TRIM ACTIVATION		FDG	FDF		FDH		11111111	
T39	SW., TRIM NORMAL < 2 EAK	14113	FDGA	FDG	1				
T39	SW., EMER DISCONNECT < 2 EAK	14116	FDGB	FDG	1				
T39	ALTERNATE TRIM ACTIVATION		FDH	FDF		K FDG		AAAAAAAA	
T39	SW., TRIM ALTERNATE < 2 EAK	14115	FDHA	FDH	A				
T39	WEIGHT REDUCTION		FE	F		K BAA		00000010	
T39	SPEED REDUCTION ATTENUATION		FF	F				11111111	
T39	SPEED REDUCTION ATTN		FF	GA				11111111	
T39	SPEED BRAKE POSITIONING		FFB	FF				00000100	
T39	SPEED BRAKE DOOR ASSY	14810	FFBA	FFB	1				
T39	SKIN	14813	FFBB	FFB	0				
T39	HINGE	14815	FFBC	FFB	A				
T39	SUPPORT	14816	FFBD	FFB	1				
T39	CIRCUIT BREAKER	42152	FFBE	FFB	1				
T39	SPEED BRAKE ACTUATION		FFC	FFB				AAAAAAAA	
T39	ACTUATOR < 2 EAK	14821	FFCA	FFC	1				
T39	VALVE, SELECTOR	14823	FFCB	FFC	5				
T39	VALVE, SHUTTLE < 2 EAK	14825	FFCC	FFC	1				
T39	SELECTOR SWITCH	14831	FFCD	FFC	A				
T39	NORMAL ACTUATION		FFD	FFC		FFE		11111111	
T39	VALVE, THERMAL RELIEF	14822	FFDA	FFD	2				
T39	VALVE, CHECK	14827	FFDB	FFD	0				
T39	ALTERNATE ACTUATION		FFE	FFC		K FFD		AAAAAAAA	
T39	VALVE, CHECK	14827	FFEA	FFE	0				
T39	VALVE, EMER SELECTOR	14824	FFEB	FFE	A				
T39	PROVISORY ATTENUATION		FFF	FFB		T		AAAAAAAA	
T39	EMER RETRACT		FFG	FFF		K FFC		AAAAAAAA	
T39	VALVE, DUMP	14828	FFGA	FFG	A				
T39	SW., EMER DUMP	9914H	FFGB	FFG	A				
T39	SW., SELECTOR	14831	FFGC	FFG	A				
T39	SPEED BRAKE WARNING		FFH	FFB		K FFC		00000000	

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 12345678901234567890123456789012345678901234567890123456789012345678901234567890

T39 SW.,LIMIT	14832	FFHA	FFH	A	
T39 LIGHT,WARNING	14834	FFHB	FFH	1	
T39 LIGHT,PANEL	44241	FFHC	FFH	1	
T39 LIGHT,MASTER	44242	FFHD	FFH	1	
T39 LIGHT,INDICATOR	44243	FFHE	FFH	1	
T39 RLY,TEST	44244	FFHF	FFH	0	
T39 GUST LOCK		FG	F		00000000
T39 GUST LOCK ASSY	14911	FGA	FG	1	
T39 TELEFLEX CABLE SYSTEM	14913	FGB	FG	1	
T39 GROUND CONTROL		G			AAAAAAAAA
T39 SPFED CONTROL		GA	G	C	000000A0
T39 SPEED CONTROL		GA	GAX		50000005
T39 WHEEL BRAKING %EA OF 2<		GAA	GA		AAAAAAAAA
T39 WHEEL BRAKING %EA OF 2<		GAA	GAD		AAAAAAAAA
T39 ROD ASSY	13611	GAAA	GAA	8	
T39 BELL CRANK	13612	GAAB	GAA	A	
T39 BRAKE ASSY	13621	GAAC	GAA	8	
T39 BRAKE CONTROL VALVE	13622	GAAD	GAA	A	
T39 SHUTTLE VALVE	13623	GAAE	GAA	5	
T39 PRESSURE PLATE	13626	GAAF	GAA	A	
T39 BRAKE DISC	13627	GAAG	GAA	A	
T39 BACK PLATE	13628	GAAH	GAA	A	
T39 DUAL VALVE ASSY	13647	GAAJ	GAA	1	
T39 PEDAL %2EA<	14521	GAAK	GAA	1	
T39 PARKING BRAKE		GAC	GAA		00000000
T39 PARKING BRAKE	13640	GACA	GAC	8	
T39 TELEFLEX ASSY	13642	GACB	GAC	8	
T39 DUAL VALVE ASSY	13647	GACC	GAC	8	
T39 DIFFERENTIAL BRAKING		GAD	GB	K GBA	0A00000A
T39 ATTENUATION		GAX	G		11111111
T39 DIRECTIONAL CONTROL		GB	G		11000011
T39 NOSE WHEEL STEERING		GBA	GB	GAD	11000011
T39 NOSE WHEEL STEERING		GBA	GBJ		FAAAAAAAAA
T39 LINKAGE %2EA<	13511	GBAA	GBA	1	
T39 BELLCRANK %2EA<	13512	GBAB	GBA	1	
T39 ROD ASSY %2EA<	13513	GBAC	GBA	1	
T39 LOCKING LEVER	13514	GBAD	GBA	0	
T39 PEDAL %4EA<	14521	GBAE	GBA	1	
T39 STEERING ACTUATION		GBB	GBA		AAAAAAAAA
T39 CONTROL VALVE	13521	GBBA	GBB	A	
T39 SWIVEL VALVE	13522	GBBB	GBB	5	
T39 ACTUATING CYLINDER	13523	GBBC	GBB	8	
T39 CONTROL VALVE FILTER	13524	GBBD	GBB	0	
T39 CHECK/THERMAL RELIEF VALV	13525	GBBE	GBB	1	
T39 STEERING CONTROL		GRC	GBA		AAAAAAAAA
T39 CIRCUIT BREAKER	42152	GBCA	GBC	1	
T39 LOAD SWITCH MLG EA. OF 2	13244	GBCB	GBC	1	
T39 MAIN CONTROL		GBD	GBC	K GBF	AAAAAAAAA
T39 LOAD SWITCH NOSE GEAR	13443	GBDA	GBD	A	
T39 CONT WHEEL/COLUMN PILOT	14111	GBDB	GBD	1	

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1234567890123456789012345678901234567890123456789012345678901234567890
T39 CONT WHEEL/COLUMN COPILOT 14112 GBDC GBD 1
T39 NORMAL SELECT GBE GBD AAAAAAAAAA
T39 PEDESTAL 12116 GBEA GBE 1
T39 STANDBY CONTROL GBF GPC K GHD AAAAAAAAAA
T39 AUTO SELECT GBG GBF GBH 111111111
T39 CONTROL BOX 13531 GBGA GBG 8
T39 FEED BACK POT 13532 GBGB GBG A
T39 COMMAND POT 13533 GBGC GBG A
T39 CONTROL RELAY 13534 GBGD GBG 5
T39 HOLD RELAY 13535 GBGE GBG 5
T39 MANUAL SELECT GBH GPF K GHF AAAAAAAAAA
T39 PEDESTAL 12116 GBHA GBH 1
T39 STEERING INDICATION GBJ GBH 333333333
T39 MONITOR SWITCH 13536 GBJA GBJ 0
T39 LIGHT PANEL 44241 GBJB GBJ 1
T39 LIGHT MASTER 44242 GBJC GBJ 1
T39 LIGHT INDICATOR 44243 GBJD GBJ 1
T39 HYDRAULIC DIST GCA G S110000011
T39 HYDRAULIC DIST GCA G C S000000080
T39 HYDRAULIC DIST GCA GAA FAAAAAAAAA
T39 THERMAL RELIEF VALVE 13624 GCAA GCA 1
T39 EMERGENCY PRESSURE GCB GCA K GCC AAAAAAAAAA
T39 RESERVOIR 13625 GCBA GCB A
T39 EMERGENCY BRAKE 13630 GCBB GCB 8
T39 HANDLE ASSY 13631 GCBC GCB 8
T39 TELEFLEX ASSY 13632 GCBD GCB 8
T39 NORMAL DIST GCC GBB FAAAAAAAAA
T39 NORMAL DIST GCC GCA GCH 111111111
T39 MISSION SUPPORT M AAAAAAAAAA
T39 SUPPORT EQUIPMENT MA M C00000000
T39 FLOOR 1211A MA MA 0
T39 INERTIA REEL 1211B MAB MA 0
T39 UPHOLSTERY 1211D MAC MA 0
T39 SEAT COCKPIT 12130 MAL MA 1
T39 WELD ASSY 12132 MAM MA 1
T39 SAFETY BELT 12133 MAN MA 1
T39 BAGGAGE COMPARTMENT GEN 12200 MAP MA C
T39 CABIN COMPARTMENT GEN 12300 MAQ MA C
T39 SEAT CABIN 12330 MAR MA 0
T39 SEAT PAN 12332 MAS MA 0
T39 SEAT BELT 12336 MAT MA 1
T39 CHURCH KEY 9912X MAV MA A
T39 LANDING GEAR N AAAAAAAAAA
T39 EXTEND GEAR NA N 0000000A0
T39 EXTEND MAIN GEAR NAA NA AAAAAAAAAA
T39 NORMAL CONTROL NAB NA NAC 111111111
T39 INST PANEL PILOT INBRD 12112 NABA NAB 1
T39 GEAR AND DOOR CONT VALVE 13111 NABH NAB A
T39 GEAR CONTROL HANDLE 13112 NABC NAB A
T39 DOWNLOCK SWITCH 2EA 13242 NABD NAB A

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12345678901234567890123456789012345678901234567890123456789012345678901234567890
T39 AFT RELAY PANEL 9913A NABE NAB 1
T39 CIRCUIT BREAKER 42152 NABG NAB 1
T39 CONTROL UNLOCK BACKUP NAC NA K NAB AAAAAAAAAA
T39 CONTROL UNLOCK BACKUP NAC NAD K NAE AAAAAAAAAA
T39 CONTROL UNLOCK BACKUP NAC NBB K NBC AAAAAAAAAA
T39 DOOR ACTUATE EA OF 2 NAD NAA AAAAAAAAAA
T39 TORQUE TUBE FAIRING DOOR 11242 NADA NAD A
T39 LOCK ROLLER FAIRING DOOR 11246 NADB NAD A
T39 LOCK ASSY STRUT DOOR 1125A NADC NAD A
T39 LOCK ROLLER STRUT DOOR 11253 NADD NAD A
T39 DOOR UNLOCK AND OPEN NAE NAD NAC 111111111
T39 DOOR ACTUATING CYLINDER 13234 NAEA NAE A
T39 SHUTTLE VALVE 13235 NAEB NAE 1
T39 CHECK VALVE 13236 NAEC NAE 1
T39 RESTRICTOR 13237 NAED NAE 1
T39 DOOR OPERATE NAF NAD AAAAAAAAAA
T39 MAIN GEAR DOOR ASSY 11240 NAFA NAF 8
T39 ROD 11243 NAFB NAF A
T39 HINGE FORWARD 11244 NAFC NAF 1
T39 HINGE AFT 11245 NAFD NAF 1
T39 MG STRUT DOOR ASSY 11250 NAFE NAF 8
T39 DOOR ATTACHING 1321D NAFF NAF 1
T39 TORQUE LINK 13214 NAFG NAF 8
T39 EMERGENCY CONTROL NAG NAC AAAAAAAAAA
T39 EMERGENCY CONTROL NAG NAK AAAAAAAAAA
T39 EMERGENCY CONTROL NAG NAN AAAAAAAAAA
T39 EMERGENCY CONTROL NAG NBB AAAAAAAAAA
T39 INSTR PANEL PLT OUTBRD 12111 NAGA NAG 1
T39 INSTR PANEL PLT INBRD 12112 NAGB NAG 1
T39 EMERGENCY SYSTEM 1314C NAGC NAG 0
T39 HANDLE ASSY 13141 NAGD NAG A
T39 DOOR EMERG UNLOCK AND OPEN NAH NAC AAAAAAAAAA
T39 CABLE ASSY 13142 NAHA NAH A
T39 LINKAGE 13145 NAHB NAH A
T39 LG DUMP VALVE DOOR 13148 NAHC NAH 5
T39 GEAR UNLOCK EA OF 2 NAJ NAA AAAAAAAAAA
T39 UPLOCK MECHANISM 13216 NAJA NAJ A
T39 UPLOCK RELEASE BACKUP NAK NAJ K NAL AAAAAAAAAA
T39 UPLOCK RELEASE NAL NAJ NAK 111111111
T39 UPLOCK ACTUATING CYLINDER 13232 NALA NAL 3
T39 ACTUATOR PIN 1321F NALB NAL A
T39 DOOR UPLOCK SWITCH 13243 NALC NAL A
T39 GEAR ACTUATE EA OF 2 NAM NAA AAAAAAAAAA
T39 DOWNLOCK MECHANISM 13217 NAMA NAM A
T39 STRUT ASSY 13212 NAMB NAM 8
T39 SIDE BRACE 13213 NAMC NAM A
T39 LINKAGE 13215 NAMD NAM A
T39 ROTATE AND LOCK BACKUP NAN NAP K NAP AAAAAAAAAA
T39 ROTATE AND LOCK NAP NAN 111111111
T39 ACTUATING CYLINDER 13231 NAPA NAP A
    
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1234567890123456789012345678901234567890123456789012345678901234567890				
T39	DOWNLOCK ACTUATING CYL	13233	NAPB	NAP A
T39	CHECK VALVE	13236	NAPC	NAP 1
T39	DOOR CLOSE AND LOCK		NAQ	NAP 000000000
T39	TORQUE TUBE	11242	NAQA	NAQ A
T39	LOCK ROLLER	11246	NAQB	NAQ A
T39	DOOR ACTUATING CYLINDER	13234	NAQC	NAQ A
T39	DOWNLOCK SWITCH	13242	NAQD	NAQ A
T39	EMERGENCY GEAR RELEASE		NAR	NAC AAAAAAAAAA
T39	EMERGENCY GEAR RELEASE		NAR	NAK AAAAAAAAAA
T39	CARLE ASSY	13142	NARA	NAR A
T39	LINKAGE	13145	NARB	NAR A
T39	LG DUMP VALVE GEAR	13148	NARC	NAR 5
T39	GEAR FREE FALL AND LOCK		NAS	NAC AAAAAAAAAA
T39	GEAR FREE FALL AND LOCK		NAS	NAK AAAAAAAAAA
T39	GEAR FREE FALL AND LOCK		NAS	NAN AAAAAAAAAA
T39	GEAR FREE FALL AND LOCK		NAS	NBG AAAAAAAAAA
T39	BUNGEE 2EA	13146	NASA	NAS 5
T39	BUNGEE NOSE	13414	NASB	NAS 5
T39	EXTEND INDICATION		NAT	NAG AAAAAAAAAA
T39	WARNING SYSTEM	13120	NATA	NAT 0
T39	POSITION INDICATOR LIGHT	1312A	NATB	NAT 1
T39	CONTROL HANDLE LIGHT	1312B	NATC	NAT 1
T39	DIMMING RESISTOR	1312C	NATD	NAT 0
T39	HORN	13121	NATE	NAT 1
T39	CUTOUT SWITCH	13122	NATF	NAT 0
T39	THROTTLE POSITION SWITCH	13123	NATG	NAT 0
T39	RELAY	13124	NATH	NAT 0
T39	SIGNAL GENERATOR	13125	NATJ	NAT 0
T39	RECTIFIER	13126	NATK	NAT 0
T39	CIRCUIT BREAKER COCKPIT	13128	NATL	NAT 1
T39	DOWNLOCK SWITCH 2EA	13242	NATM	NAT A
T39	DOOR UPLOCK SWITCH 2EA	13243	NATN	NAT 1
T39	DOWNLOCK SWITCH	13442	NATP	NAT A
T39	AIRSPPEED ALT WARN SWITCH	51246	NATQ	NAT 0
T39	EXTEND NOSE GEAR		NBA	NA AAAAAAAAAA
T39	GEAR AND DOOR UNLOCK		NBB	NBA AAAAAAAAAA
T39	LOCK NG DOOR	11227	NBBA	NBB A
T39	UPLOCK MECHANISM	13410	NBBB	NBB A
T39	GEAR AND DOOR RELEASE		NBC	NBB NAC 111111111
T39	TORQUE LINK	13416	NBCA	NBC A
T39	UPLOCK ACTUATING CYLINDER	13432	NBCB	NBC A
T39	GEAR ACTUATE		NBF	NBA AAAAAAAAAA
T39	DOWN LOCK MECHANISM	1341E	NBFA	NBF A
T39	AXLE CENTERING BLOCK ASSY	1341C	NBFB	NBF 1
T39	STRUT ASSY	13412	NBFC	NBF 8
T39	DRAG BRACE	13413	NBFD	NBF A
T39	TRUNNION	13415	NBFE	NBF A
T39	TORQUE LINK	13416	NBFF	NBF A
T39	GEAR ROTATE AND LOCK BACKUP		NBG	NBF K NBH AAAAAAAAAA
T39	GEAR ROTATE AND LOCK		NBH	NBF NBG 111111111

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123456789012345678901234567890123456789012345678901234567890123456789012345678901234567890
T39 ACTUATING CYLINDER GEAR 13431 NBHJ NBH A
T39 DOWNLOCK ACTUATING CYL 13433 NBHK NBH A
T39 RESTRICTOR 13434 NBHL NBH 1
T39 DOOR OPEN NBJ NBF AAAAAAAAAA
T39 NOSE GEAR DOOR ASSY 11220 NBJA NRJ 1
T39 LINKAGE 1122A NBJB NRJ 2
T39 HINGE FORWARD 11222 NBJC NRJ 0
T39 HINGE AFT 11223 NBJD NRJ 0
T39 RETRACT GEAR NC NX 010000000
T39 RETRACT CONTROL NCA NC AAAAAAAAAA
T39 INST PANEL PILOT INBOARD 12112 NCA NCA 1
T39 GEAR AND DOOR CONT VALVE 13111 NCAB NCA A
T39 GEAR CONTROL HANDLE 13112 NCAC NCA A
T39 UPLOCK SWITCH 2EA 13241 NCAD NCA A
T39 LOAD SWITCH 2EA 13244 NCAE NCA 1
T39 UPLOCK SWITCH NOSE 13441 NCAF NCA A
T39 CIRCUIT BREAKER 42152 NCAG NCA 1
T39 AFT RELAY PANEL 9913A NCAH NCA A
T39 MLG RETRACT EA OF 2 NCB NC AAAAAAAAAA
T39 DOOR OPERATION NCD NCB AAAAAAAAAA
T39 MAIN GEAR DOOR ASSY 11240 NCDA NCD 1
T39 TORQUE TUBE 11242 NCDB NCD A
T39 ROD 11243 NCDC NCD A
T39 HINGE FORWARD 11244 NCDD NCD A
T39 HINGE AFT 11245 NCDE NCD A
T39 LOCK ROLLER 11246 NCDF NCD A
T39 DOOR ACTUATING CYLINDER 13234 NCDG NCD 8
T39 SHUTTLE VALVE 13235 NCDH NCD 1
T39 CHECK VALVE 13236 NCDJ NCD 1
T39 RESTRICTOR 13237 NCDK NCD 1
T39 MAIN GEAR STRUT DOOR ASSY 11250 NCDL NCD 1
T39 LOCK ASSY 1125A NCDM NCD A
T39 LOCK ROLLER 11253 NCDN NCD A
T39 DOOR ATTACHING 1321D NCDP NCD A
T39 GEAR LOCK RELEASE NCE NCB AAAAAAAAAA
T39 TORQUE LINK 13214 NCEA NCE A
T39 DOWNLOCK MECHANISM 13217 NCEB NCE A
T39 DOWNLOCK ACTUATING CYL 13233 NCEC NCE A
T39 CHECK VALVE 13236 NCED NCE 1
T39 DOOR UPLOCK SWITCH 13243 NCEE NCE A
T39 GEAR UP AND LOCK NCF NCB AAAAAAAAAA
T39 ACTUATOR PIN 1321F NCFA NCF A
T39 STRUT ASSY 13212 NCFB NCF 8
T39 SIDE BRACE 13213 NCFD NCF A
T39 TORQUE LINK 13214 NCFD NCF A
T39 LINKAGE 13215 NCFE NCF A
T39 UPLOCK MECHANISM 13216 NCFE NCF A
T39 ACTUATING CYLINDER 13231 NCFG NCF A
T39 UPLOCK ACTUATING CYLINDER 13232 NCFH NCF A
T39 RESTRICTOR 13237 NCFJ NCF 1
    
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00000000111111112222222222333333333344444444445555555555666666666677777777778
12345678901234567890123456789012345678901234567890123456789012345678901234567890
T39 ESSENTIAL AC DISTRIBUTION UAB HSJ AAAAAAAAAA
T39 ESSENTIAL AC DISTRIBUTION UAB BSY AAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CFA AAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CFL FAAAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CHBA FAAAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CHBR FAAAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CHC AAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CHEB FAAAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CJ FAAAAAAAAAAA
T39 ESSENTIAL AC BUS UAB CL AAAAAAAAAA
T39 ESSENTIAL AC DIST UAB DAA AAAAAAAAAA
T39 ESSENTIAL AC DIST UAB EAE 111111111
T39 ESSENTIAL AC DIST UAB ECR AAAAAAAAAA
T39 ESSENTIAL AC DIST UAB EEG 555555555
T39 ESSENTIAL AC DIST UAB UHG 111111111
T39 NORMAL FEED UAC UAB UAD 111111111
T39 INVERTER CHANGOVER CONTROL 42232 UAC UAC 1
T39 RELAY AC TRANSFER 42234 UACB UAC 1
T39 EMERGENCY FEED UAD UAB K UAC AAAAAAAAAA
T39 INVERTER CHANGE OVER CONTR 42232 UADA UAD A
T39 RELAY AC TRANSFER 42234 UADB UAD A
T39 MAIN INVERTER SYSTEM UAE UAA AAAAAAAAAA
T39 MAIN INVERTER SYSTEM UAE UAC AAAAAAAAAA
T39 INVERTER MAIN 42221 UAEA UAC A
T39 SWITCH INVERTER 42224 UAEB UAF A
T39 CKT BKRS MAIN INV 42225 UAEC UAF A
T39 RELAY MAIN INV POWER 42238 UAED UAF A
T39 STANDBY INVERTER SYSTEM UAF UAD AAAAAAAAAA
T39 INVERTER STANDBY 42222 UAFA UAF A
T39 CKT BKRS,STBY INV 42225 UAFB UAF A
T39 SWITCH INVERTER 42224 UAFC UAF A
T39 POWER WARNINGS UAG UAD 000000000
T39 CAUTION LIGHT PANEL 44241 UAGA UAG 1
T39 MASTER CAUTION LIGHT 44242 UAGB UAG 0
T39 CAUTION LIGHT, INV FAIL 44243 UAGC UAG 1
T39 CAUTION LIGHT, INST PWRUFF 44243 UAGD UAG 1
T39 RELAY, TEST 44244 UAGE UAG 0
T39 ESSENTIAL DC DISTRIBUTION UDA BAM AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BAP AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BAZC AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BAZD AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BAZK AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BBA AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BPH SAAAAAAAAAAA
T39 ESSENTIAL DC DIST UDA BPC FAAAAAAAAAAA
T39 ESSENTIAL DC DIST UDA BPD FAAAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BPH AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BSE AAAAAAAAAA
T39 ESSENTIAL DC DISTRIBUTION UDA BSN FAAAAAAAAAAA

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123456789012345678901234567890123456789012345678901234567890123456789012345678901234567890							
T39 ESSENTIAL DC DISTRIBUTION	UDA			BSS		FAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			BST		FAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			BSU		FAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			BSV		FAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			BSW		AAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			BSZ		FAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			BTA		AAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CA		AAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CB		AAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CC		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			CE		S008888800	
T39 ESSENTIAL DC BUS	UDA			CFA		FAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CFL		FAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CG		FAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CHCY		AAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CHEA		FAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CHF		AAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CJ		FAAAAAAAAA	
T39 ESSENTIAL DC BUS	UDA			CL		AAAAAAAAAA	
T39 ESSENTIAL DC	UDA			DAB		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			DAV		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			DBN		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			DC		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			DEA		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EAE		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EAN		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EAP		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EAQ		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EBA		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EBB		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			ECA		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EDA		FAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			EEG		FAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			FAA		00AAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			FBD		AAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			FCE		AAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			FDG		AAAAAAAAAA	
T39 ESSENTIAL DC DISTRIBUTION	UDA			FFB		AAAAAAAAAA	
T39 ESSENTIAL DC	UDA			GBC		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			NAB		FAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			NAT		FAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			NCA		FAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			NCM		FAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			UAF		FAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			UHB		AAAAAAAAAA	
T39 ESSENTIAL DC DIST	UDA			UHG		F111111111	
T39 CKT BKRS ESSENTIAL BUS	42152	UDAA		UDA		1	
T39 BOX POWER DIST	42157	UDAB		UDA		1	
T39 BOX RELAY	42158	UDAC		UDA		1	
T39 TERMINAL STRIP	42150	UDAD		UDA		0	

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12345678901234567890123456789012345678901234567890123456789012345678901234567890
T39 SECONDARY DC BUS UDB CFBB AAAAAAAAAA
T39 SECONDARY DC BUS UDB CFDB FAAAAAAAAA
T39 SECONDARY DC BUS UDB CFEC FAAAAAAAAA
T39 SECONDARY DC BUS UDB CHBC FAAAAAAAAA
T39 SECONDARY DC BUS UDB CHED FAAAAAAAAA
T39 SECONDARY DC DIST VIBRATE UDB DA F00000000
T39 SECONDARY DC DIST UDB DBQ AAAAAAAAAA
T39 SECONDARY DC DIST UDB EAE 111111111
T39 SECONDARY DC DIST UDB EAF AAAAAAAAAA
T39 SECONDARY DC DIST UDB EBB AAAAAAAAAA
T39 SECONDARY DC DIST UDB ECP AAAAAAAAAA
T39 CKT BKRS, SEC BUS 42152 UDDBA UDB A
T39 BOX POWER DIST 42157 UDDB UDB 1
T39 BOX RELAY 42158 UDBC UDB 1
T39 RFLAY SECONDARY BUS 99428 UDRD UDR A
T39 TERMINAL STRIP 42150 UDBE UDB 1
T39 STARTER BUS POWER DIST UDC BAN AAAAAAAAAA
T39 PARALELLING BUS UDD UAA SAAAAAAAAA
T39 PARALELLING BUS UDD UAB S111111111
T39 PARALELLING BUS UDD UAE FAAAAAAAAA
T39 PARALELLING BUS UDD UDB FAAAAAAAAA
T39 PARALELLING BUS UDD UDH AAAAAAAAAA
T39 PARALELLING BUS UDD UHH AAAAAAAAAA
T39 TERMINAL STRIP 42150 UDDA UDD C
T39 BOX POWER DIST 42157 UDDB UDD 1
T39 BOX RELAY 42158 UDDC UDD 1
T39 BATTERY BUS UDE DDB AAAAAAAAAA
T39 BATTERY BUS UDE EAD AAAAAAAAAA
T39 BATTERY BUS UDE UAB K UDH SAAAAAAAAA
T39 BATTERY BUS UDE UDJ SAAAAAAAAA
T39 BATTERY BUS UDE UDK FAAAAAAAAA
T39 BATTERY BUS UDE UDL FAAAAAAAAA
T39 BATTERY BUS UDE UDM FAAAAAAAAA
T39 TERMINAL STRIP 42150 UDEA UDE 1
T39 BOX POWER DIST 42157 UDEB UDE 1
T39 BOX RELAY 42158 UDEC UDE 1
T39 EXTERNAL START POWER UDF UDC 100000000
T39 RECEPTACLE EXT START 42154 UDFA UDF A
T39 MICROSWITCH 42155 UDFB UDF 1
T39 BATTERY START POWER UDG UDC 100000000
T39 RELAY BATTERY START 42171 UDGA UDG A
T39 NORMAL OPERATION UDH UDA UDJ 111111111
T39 RELAY ESSENTIAL BUS 42177 UDHA UDH A
T39 BACKUP OPERATION UDJ UDA K UDH AAAAAAAAAA
T39 RELAY ESS BUS EMER K512 9942A UDJA UDJ A
T39 MASTER SWITCH 42156 UDJB UDJ A
T39 AUTO SWITCHING UDK UDL 111111111
T39 RELAY GEN MONITOR 4217C UDKA UDK A
T39 BATTERY SWITCH 4211B UDKB UDK A
T39 MANUAL SWITCHING UDL UDJ K UDK AAAAAAAAAA
    
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FLIGHT SAFETY PREDICTION TECHNIQUE

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1234567890123456789012345678901234567890123456789012345678901234567890						
T39 BATTERY SWITCH	42118	UDLA	UDL	A		
T39 BATTERY PARALLELING		UDM	UDD		0000C0000	
T39 RELAY BATTERY	42176	UDMA	UDM	A		
T39 GENERATOR STATUS		UDN	UDJ		0000C0000	
T39 VOLTMETER	4213C	UDNA	UDN	1		
T39 GEN OFF WARN LIGHT	44243	UDNB	UDN	1		
T39 CAUTION LIGHT PANEL	44241	UDNC	UDN	1		
T39 MASTER CAUTION LIGHT	44242	UDND	UDN	0		
T39 RELAY TEST	44244	UDNE	UDN	0		
T39 GEN LIGHT CKT BKR	42152	UDNF	UDN	1		
T39 LOAD SWITCH	4215F	UDNG	UDN	0		
T39 VOLTMETER TEST SWITCH	4213A	UDNH	UDN	1		
T39 LOADMETER	4213B	UDNJ	UDN	1		
T39 CKT BKR VOLTMETER	42152	UDNK	UDN	1		
T39 EXTERNAL UTILITY POWER		UDP	UDD		000000000	
T39 PLUG DISCONNECT	4215C	UDPA	UDP	0		
T39 RECEPTACLE EXT UTIL	42153	UDPB	UDP	8		
T39 RELAY EXT PWR CUTOFF	42172	UDPC	UDP	A		
T39 RELAY EXT PWR MONITOR	42175	UDPD	UDP	A		
T39 LEFT GENERATOR SYSTEM		LUDQ	UDR		AAAAAAAAA	
T39 LEFT GENERATOR SYSTEM		LUDQ	UDD	KRUDQ	AAAAAAAAA	
T39 RIGHT GENERATOR SYSTEM		RUDQ	UDB		AAAAAAAAA	
T39 RIGHT GENERATOR SYSTEM		RUDQ	UDD	KLUDQ	AAAAAAAAA	
T39 STARTER GENERATOR	42131	LUDQA	LUDQ	A		
T39 STARTER GENERATOR	42131	RUDQA	RUDQ	A		
T39 ADAPTER QAD	42132	LUDQB	LUDQ	A		
T39 ADAPTER QAD	42132	RUDQB	RUDQ	A		
T39 DUCT FLEX COOLING	42135	LUDQC	LUDQ	1		
T39 DUCT FLEX COOLING	42135	RUDQC	RUDQ	1		
T39 VOLTAGE REGULATOR	42136	LUDQD	LUDQ	A		
T39 VOLTAGE REGULATOR	42136	RUDQD	RUDQ	A		
T39 RELAY GEN MONITOR	4217C	LUDQE	LUDQ	A		
T39 RELAY GEN MONITOR	4217C	RUDQE	RUDQ	A		
T39 RELAY GEN FIELD CONTROL	4217D	LUDQF	LUDQ	A		
T39 RELAY GEN FIELD CONTROL	4217D	RUDQF	RUDQ	A		
T39 CKT BKRS	42152	LUDQG	LUDQ	A		
T39 CKT BKRS	42152	RUDQG	RUDQ	A		
T39 MASTER SWITCH	42156	LUDQH	LUDQ	A		
T39 MASTER SWITCH	42156	RUDQH	RUDQ	A		
T39 REVERSE CURRENT RELAY	9942C	LUDQJ	LUDQ	A		
T39 REVERSE CURRENT RELAY	9942C	RUDQJ	RUDQ	A		
T39 GEN SWITCH	9942D	LUDQK	LUDQ	A		
T39 GEN SWITCH	9942D	RUDQK	RUDQ	A		
T39 BATTERY SYSTEM NO.1		LUDR	UDE		555555555	
T39 BATTERY SYSTEM NO.1		LUDR	UDG		AAAAAAAAA	
T39 BATTERY SYSTEM NO.2		RUDR	UDE		555555555	
T39 BATTERY SYSTEM NO.2		RUDR	UDG		AAAAAAAAA	
T39 BATTERY	42111	LUDRA	LUDR	8		
T39 BATTERY	42111	RUDRA	RUOR	8		
T39 QUICK DISCONNECT	42114	LUDRB	LUDR	A		

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T39 QUICK DISCONNECT	42114	RUDRB	RUDK	A
T39 CABLE	42115	LUORC	LUOR	8
T39 CABLE	42115	RUDRC	RUDR	8
T39 SUMP JAR	42116	LUORD	LUOR	1
T39 SUMP JAR	42116	RUDRD	RUDR	1
T39 VENT AND DRAIN HOSE	42118	LUORE	LUOR	0
T39 VENT AND DRAIN HOSE	42118	RUDRE	RUDR	0
T39 NORMAL HYD DISTRIBUTION		UHA	FFD	AAAAAAAAA
T39 NORMAL HYDRAULIC DIST		UHA	GCC	UHD 111111111
T39 NORMAL HYDRAULIC DIST		UHA	NAB	S555555555
T39 NORMAL HYDRAULIC DIST		UHA	NCA	S555555555
T39 CHECK VALVE HYDR 4FA	4531A	UHAA	UHA	1
T39 FILLER VALVE ACCUMULATOR	4521C	UHAB	UHA	1
T39 ACCUMULATOR	45211	UHAC	UHA	1
T39 RELIEF VALVE	45214	UHAD	UHA	1
T39 DUMP VALVE	45215	UHAE	UHA	0
T39 FILTER PRESSURE	45216	UHAF	UHA	1
T39 NORMAL HYD PRESSURE		UHB	UHA	AAAAAAAAA
T39 NORMAL HYD PRESSURE		UHB	UHD	FA00000000
T39 CIRCUIT BREAKER 2EA	42152	UHBA	UHB	1
T39 CHECK VALVE HYDR	4511A	UHBB	UHB	0
T39 PRESSURE SWITCH 2EA	4511F	UHBC	UHB	A
T39 MOTOR HYDRAULIC PUMP	4511L	UHBD	UHB	A
T39 MOTOR DRIVEN PUMP	45111	UHBE	UHB	8
T39 FILTER RETURN	45217	UHBF	UHB	1
T39 HYDRAULIC PUMP SWITCH	9945A	UHBG	UHB	A
T39 HYDRAULIC SUPPLY		UHC	UHB	AAAAAAAAA
T39 CHECK VALVE AIR	4511R	UHCA	UHC	1
T39 DRAIN VALVE AIR TANK	4511C	UHCB	UHC	0
T39 FILTER AIR	4511D	UHCC	UHC	0
T39 COUPLING EXTERNAL HYD	4511G	UHCD	UHC	0
T39 COUPLING EXTERNAL AIR	4511J	UHCH	UHC	0
T39 CAP AIR COUPLING	4511K	UHCL	UHC	0
T39 PRESSURE REGULATOR	45114	UHCK	UHC	1
T39 RESERVOIR ASSY	45115	UHCL	UHC	8
T39 TANK AIR	45116	UHCM	UHC	1
T39 FILLER CAP RESERVOIR	45117	UHCN	UHC	0
T39 DRAIN VALVE RESERVOIR	45118	UHCP	UHC	1
T39 AUXILLIARY HYD DISTRIBUTION		UHD	FFE	AAAAAAAAA
T39 AUXILIARY HYDRAULIC DIST		UHD	GCC	K UHA AAAAAAAAAA
T39 CHECK VALVE NORMAL 4 FA	4511A	UHDA	UHD	1
T39 ACCUMULATOR AUX	45311	UHDB	UHD	A
T39 FILLER VALVE ACCUM AUX	45313	UHDC	UHD	8
T39 RELIEF VALVE AUX	45316	UHDD	UHD	1
T39 DUMP VALVE AUX	45317	UHDE	UHD	0
T39 SHUT OFF VALVE AUX	45318	UHDF	UHD	A
T39 CKT BKR	42152	UHDG	UHD	0
T39 PRESSURE INDICATION		UHG	UHD	111111111
T39 CIRCUIT BREAKER	42225	UHGA	UHG	1
T39 LIGHT PANEL CAUTION	44241	UHGB	UHG	1

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T39 LIGHT MASTER CAUTION 44242 UHGC UHG 1
T39 LIGHT INDICATOR CAUTION 44243 UHGD UHG 1
T39 GAGE PRESSURE 4521A UHGE UHG 1
T39 GAGE ACCUMULATOR NORM 4521B UHGF UHG 0
T39 PRESSURE TRANSMITTER NORM 45213 UHGG UHG 1
T39 GAGE ACCUMULATOR AUX 45312 UHGH UHG 0
T39 PRESSURE TRANSMITTER AUX 45315 UHGJ UHG 0
T39 PRESSURE SNUBBER 45212 UHGK UHG 0
T39 PRESSURE SNUBBER AUX 45314 UHGL UHG 0

CARD COUNT IS 00001808. CARDS WITH ERRORS 0000000

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