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ABSTRACT

Hydrostatic pressure tests in the NRL 8000 psi pressure chamber were made on four Massa TR-11C variable reluctance, magnetic field transducers. Some of the effects of hydrostatic pressure on structural strength and functional behavior of these transducers are discussed.

PROBLEM STATUS

This is an interim report on this phase of the problem; work on other phases is continuing.

AUTHORIZATION

NRL Problem S02-06 Task 8047 BuShips No. S-F001-03-04

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MASSA TR-11C TRANSDUCERS -HYDROSTATIC PRESSURE TESTS

INTRODUCTION

Four magnetic field transducers, designated TR-11C, Nos. 1, 2, 3 and 4, of the variable reluctance type manufactured by The Massa Division of Cohu Electronics Corporation for Project Artemis were tested in the NRL 8000 psi hydrostatic pressure chamber. The principal purpose was to determine the structural strength of the box walls and any change in functional behavior that might be observed. These transducers are different from the TR-11B series with hollow plates in that the plates forming the box enclosure were rabbeted to provide a structural support of each plate edge onto each adjacent plate edge. This change avoids reliance on the strength of the adhesive bonds only for structural support between some of the plate edges as was the case in the TR-11B series of transducers.

TEST PROCEDURE

In preparing the Massa units for the pressure test a rubber molding with a brass sleeve insert was molded onto the electrical cable of each transducer. This molding was placed about six feet from the free end of the cable. The molding served the purpose of the stuffing in a regular stuffing gland to provide a watertight seal at the cable entry of the pressure chamber. The

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transducers were tested separately. Two of them, Nos. 1 and 3, were subjected to pressures exceeding the plate strength of the transducer walls and the other two, Nos. 2 and 4, were pressure cycled. Number 2 was pressure cycled 45 times between 250 and 2000 psi. Number 4 was cycled 50 times over each of four ranges between 200 psi and 1700, 1800, 1900 and 2000 psi. The average period of one cycle was about four minutes.

The effects of hydrostatic pressure on the structural stability and functional behavior of the transducers were detected by changes in measured impedance and frequency of the principal and parasitic resonant modes of vibration. Some structural failures were also indicated by sharp, audible sounds resulting from an implosion of the box that were readily detected by ear. At low driving levels (less than one milliampere) the impedance of the transducer was measured with a vector impedance locus plotter (VILP) which indicates the magnitude of the X and R components as a function of frequency. This instrument was used to detect parasitic resonant modes and changes in the principal modes of vibration caused by application of hydrostatic pressure. Ammeter, voltmeter and wattmeter were used to measure the transducer impedance at higher driving levels and to continuously monitor the frequency at maximum impedance during application of the hydrostatic pressure.

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RESULTS

Unlike the previous Massa transducers that have been checked at NRL under hydrostatic pressures, these four developed parasitic resonant modes of vibration on application of hydrostatic pressure. Some of these modes persisted, although, in most cases, with reduced activity, over the entire range of pressures and others disappeared. In each case at pressures between 100 and 300 psi some of these parasitic modes occurred at frequencies near the frequency of the principal resonant mode and the consequent distortion of the impedance circle plot made the usual evaluation of Q, frequency, and resistance of the principal resonant mode meaningless. The parasitic resonances occurred at a pressure of 300 psi for transducers, Nos. 1 and 2, and at 100 psi for transducers, Nos. 3 and 4. In each case these modes did not persist over the entire range of pressures. For transducer No. 1, it disappeared when the pressure was decreased to 250 psi or increased above 320 psi. For No. 2. disappearance occurred at 220 and 400 psi. For No. 4, disappearance occurred at 60 psi and above 300 psi. For transducer No. 3, the pressure was not decreased to learn the behavior of this mode for decreasing pressure, and on increasing pressure this mode only diminished in its activity and did not completely disappear until the pressure was increased above 1000 psi. In

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the case of transducers Nos. 1, 2, and 4, on which the applied pressure was reduced from higher values, these parasitic modes recurred at the same pressure as originally in the case of transducer No. 1 but recurred at 200 psi for No. 2 and 70 psi for No. 4, somewhat lower than the pressures at which these modes occurred originally.

The appearance, disappearance and changes in activity with applied hydrostatic pressure of the parasitic modes occurring off-resonance with respect to the principal mode of vibration are given in Tables I, II, III, and IV. It will be noted in each table in the column of frequency under the heading "At Principal Resonance" that the change in frequency of the principal mode of vibration with pressure alone, in the range above 500 psi, was about 1 cycle per second for each 200 psi increase in pressure when excited at the low driving level that the VILP allows. At the higher driving level (100 milliamperes), at hydrostatic pressures above 500 psi, this rate is about 2 cycles per second for each 200 psi increase in pressure. This is indicated in each table in the column of data under the heading "Frequency of Maximum Impedance" and includes the effects of both pressure and driving level on resonant frequency.

Cycling the pressure applied to transducers, Nos. 2 and 4, had no effects different from the initial application over the

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same range except that the parasitic modes associated with pressure appeared at a lower pressure level than was observed initially.

The enclosure of transducer No. 1 had a structural failure when the applied pressure reached the value of 2650 psi. The diameter of the impedance circle as measured with the VILP started to decrease at the applied pressure of 2100 psi and in effect anticipated that plate deflection was sufficient to produce very light contact with the internal moving mass but not sufficient for structural failure. The failure was indicated by several sharp, audible sounds and by complete disappearance of the principal mode of vibration as observed with the continuously monitoring instruments at the time of failure and with the VILP shortly afterwards. The VILP also showed that several higher modes of vibration developed between 1000 and 2000 cps. On reducing the pressure to atmospheric pressure, a mode of vibration appeared at 413 cps and one at 491 cps.

Visual observation of this transducer, after removing the rubber covering and removing the four hollow plates constituting four walls of the enclosure, showed all plates were permanently deflected. Two of the four plates had barely perceptible permanent deflection, the third plate had considerably more visible deflection and the fourth plate was not only deflected

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but fractured as is quite evident in the photographs of Fig. 1. The latter two plates also showed separation of the two waffle halves composing the plate, indicating failure of the adhesive bond that holds the sections together. The photograph in Fig. 2 shows the two sections composing the plate. Figure 3 shows the appearance of transducer No. 3 with the four hollow plates removed. Figure 4 is a photograph taken of one radiating end of transducer No. 3 giving an outline of the plate edges showing clearly the rabbeting of the plate edges. One air gap of transducer No. 1, as measured after removal of the enclosure plates, varied from 0.024 to 0.027 inches and the other from 0.027 to 0.031 inches.

Structural failure of the enclosure wall plates of twansducer No. 3 occurred at an applied hydrostatic pressure between 2100 and 2200 psi. As in the case of transducer No. 1, structural failure was indicated by several sharp, audible sounds and disappearance of the principal mode of vibration. Instead of stopping the test at this point, the applied pressure was increased to determine the value at which the transducer electrical cable would be extruded from the stuffing gland in the test chamber cable entry. This occurred at a pressure of 8800 psi. As was expected all four walls of the transducer enclosure were fractured. This is quite evident in the photograph of Fig. 3. The plates were forced against the springs. In this

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test (up to 8800 psi) all cement bond joints of all four side plates and cement bond joints of all springs were broken. The end plates, from which acoustic radiation is produced, did not have any visibly perceptible permanent deflection. The air gaps were fairly uniform varying from 0.023 to 0.025 inches. CONCLUSION

Although applied pressure at fracture of the transducer wall plates is an obvious upper limit of hydrostatic pressure for operation of these transducers, the limiting factor for satisfactory operation is the deflection of these plates with applied hydrostatic pressure. The deflection of the plate wall was equal to the clearance of about 1/8 of an inch between plates and the free mass inside the enclosure when the applied hydrostatic pressure was 2000 psi or higher, however, the plate deflection appears to be elastic (or reversible), judging from the cycling tests, up to 2000 psi. Firm proof of the latter supposition could be obtained only by attaching strain gauges to the box prior to the tests and before the application of the rubber covering of the element and measuring the strains. This test has not been attempted. Contact between internal and external parts produced a detectable change in the rate of decrease of frequency and diameter of the impedance circle at or near the resonance of the principal mode of vibration. This

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behavior was more pronounced in the case of transducer, No. 1, between applied pressures of 2000 and 2650 psi. Observation of the plates showed evidence that two of these plates, including one that did not fracture, deflected against the free mass structure with enough force to leave permanent depressions in the plate surfaces.

A satisfactory explanation has not been made for the development of parasitic modes of appreciable activity at or near the resonant frequency of the principal mode of vibration when the applied pressure on the transducers was between 100 and 300 psi, and the disappearance of these modes at pressures below and above this range. There was associated with this pressure range a marked change in the rate of decrease in the resonant frequency of the principal mode of vibration with applied pressure. This behavior might suggest that an elastic shift or realignment of the structural assembly, probably at the bonded joints, on application of hydrostatic pressures in this range may be the common cause.

The decrease in resonant frequency of the principal mode of vibration with increased driving level and with increasing hydrostatic pressure may be accounted for in terms of the change in negative stiffness introduced by change in the air gap flux density which is a function of air gap length. The

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effective air gap length depends upon the electrical driving level and the applied hydrostatic pressure. Computations of the change of air gap length at the 100 milliampere driving level in terms of the change in negative stiffness necessary to account for the observed change in resonant frequency of the principal mode of vibration agree favorably with the change in air gap length obtained from displacement measurements of the radiating mass of the transducer at the same driving level.

Finally, since this transducer element is designed to operate in an ambient hydrostatic pressure of 600 psig, these tests indicate a safety factor of more than three and it is believed that the units will have satisfactory life characteristics. Further tests should be performed in order to determine the cause of the parasitic resonances which are associated with ambient hydrostatic pressure. It is noted that the parasitic resonance at about 500 cps, which is also observed in air measurements, is a result of the mechanical design, employed by Massa, which omits springs on two of the four sides of the structure and allows a resonance or motion perpendicular to the motion used for radiation of acoustic energy.

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TABLE I

		Loop Dia in ohms	
	ration.	Freq.	4441
-11C, No. 1	ent amperes. mode of vib	itic Modes Loop Dia. in ohms	
lucer TR	ng curr 0 milli ncipal	Freq.	00000000000000000000000000000000000000
assa Transd	res. Drivi (Zm) was lo	Loop Dia. in ohms	
Data - M	9.5 amper pedance andwidth	Freq.	disadian disadiana di anticologia di
c Test	rent = (imum im n the b	nance Q	
Hydrostati	arizing cur ency of max well withi	circle Dia in ohms	770 570 570 570 500 500 570 770 770 770
of the	Pol t frequ	At Pri Freq. cps	tured a
Some	a his freque	Freq. of Zm in cps	456 456 456 4520 4520 4225 4225 4225 4225 4225 4114 4114 4120 4112 4112 4112 4112 4112
	E	Gauge Pressure in psi	(in air) 400 100 1100 2500 2500 5000 11000 11200 11200 11200 11200 22000 2000000

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Some of the Hydrostatic Test Data - Massa Transducer TR-11C, No. 2

TABLE II

	Loop Dia. in ohms	1905-2005-190 1901-2005-200 1901-1901-1901-1901-1901-1901-1901-19
bration	Freq. cps	
rent iamperes. mode of vi	itic Modes Loop Dia. in ohms	1 1 20 000 000 000 000 000 000 1 1
ing cur 00 mill incipal	Paras Freq. cps	++++++++++++++++++++++++++++++++++++++
eres. Driv (Z_m) was l h of the pr	Loop Dia. in ohms	20000100010000000000000000000000000000
9.5 amp mpedance bandwidt	Freq. cps	4444 96144 4994 6014 7444 7444 76 7444 76 76 76 76 76 76 76 76 76 76 76 76 76
rent = cimum i	Q	
larizing cur uency of max s well withi	ncipal Resor Circle Dia. in ohms	Impedance Ci troo Tmpedance Ci troo troo troo troo troo troo troo tro
Po at freq lency wa	At Pri Freq. cps	t t t t t t t t t t t t t t
his frequ	Freq. of Zm in cps	453 453 429 4228 4228 4228 4228 4228 4228 4228
H	Gauge Pressure in psi	(in air) (in water) 100 2200 2200 2200 2000 1000 1700 2000 20

*After 45 Pressure Cycles

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Some of the Hydrostatic Test Data - Massa Transducer TR-11C, No. 3

TABLE III

	Loop Dia. in ohms	00000000000000000000000000000000000000
bration	Freq.	a1 4551 4551 4551 4551 4551 4551 4551 45
rent iamperes. mode of vi	itic Modes Loop Dia. in ohms	 80 100 100 100 40 40 40 20 20 20 20 20 20 20 10 10 20 10 20 10 20 10 20 40 10 20 100 100 100 100 100 100 100 100
ring cur 00 mill incipal	Freq.	to 1 8800 11 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
eres. Driv (Z_m) was 1 h of the pr	Loop Dia. in ohms	40 10 10 10 10 10 10 10 10 10 10 10 10 10
9.5 amp mpedance bandwidt	Freq. cps	505 505 505 505 505 505 505 799 4986 4986 4986 4986 4986 4986 4986 49
rrent = ximum i in the	ance .	r r r r r r r r r r r r r r r r r r r
larizing cu luency of ma s well with	Circle Dia in ohms	540 575 575 576 576 576 576 560 560 600 600 600 600 520 520 520 520 520 520 520 520 520 5
Po at freg lency wa	At Pri Freq. cps	4566 4406 4406 4438 4432 4432 4432 4432 4432 4432 4432
is frequ	Freq. of Z_m in cps	453 425 425 425 4125 412 412 412 412 412 412 412 412 412 412
μŢ	Gauge Pressure in psi	(in air) (in water) 100 200 700 1000 1500 1500 2100-2200

NI	
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Some of the Hydrostatic Test Data - Massa Transducer TR-11C, No. 4

Polarizing current = 9.5 amperes. Driving current at frequency of maximum impedance (Z_m) was 100 milliamperes. This frequency was well within the bandwidth of the principal mode of vibration.

	E	14 Def					C			
rauge Pressure in psi	of Zm in cps	Freq.	Circle Dia. in ohms	e de	Freq.	Loop Dia. in ohms	Freq.	Loop Dia. in ohms	Freq.	Loop Dia. in ohms
(in air) (in water) 100 200 200 1700 1700 1700(1) 1800(1) 1800(2) 1900(2) 1900(2) 1900(2) 1900(2) 1900(2) 200(3) 200(3) 200(3) 200(3) 200(3) 200(4) 200(3) 200(4) 200(5) 100(5) (in air)(5) (1) After 50 F	Lt Cortical Li Cor	storted mpedettron to to to to to to to to to to to to to	41400 1mpedance Cil 1mpedance Cil 1mpedance Cil 14400 14400 14500 14500 14500 14400 14400 14400 14400 141000 141000 1410		498 4995 4995 4995 4995 4995 492 492 492 492 492 492 492 492 492 492	60 10 10 10 10 10 10 10 10 10 1	t 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			00000000 00000000000000000000000000000
(2) After 100	Pressure	Cycles	(5) P	Vext day						

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(3) After 150 Pressure Cycles



Fig. 1(a) - Transducer TR11-C #1











