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**SIMULATION OF TRIPLE-SPOOL TURBOFAN
ENGINE**

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**Air Force Aero Propulsion Laboratory
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April 1974

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of the major engine components. Information on setting up the Block Data and input data is given in the report. Also included is a complete program listing with a description of each subroutine and sample results.

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FOREWORD

This report was prepared in the Performance Branch (AFAPL/TBA), Turbine Engine Division, Air Force Aero Propulsion Laboratory, Wright-Patterson Air Force Base, Ohio, under Project 668A0215, with Edward K. Navvaisis as Project Engineer.

This report covers work conducted within the Performance Branch in the time period between April 1973 and January 1974.

ABSTRACT

This report describes a digital computer program entitled TRISPL. TRISPL is a computer program that simulates steady-state design and off-design performance of triple-spool turbofan engines. The program has been formulated for an engine type with two core spools and one fan spool but can easily be modified for other engine types (two fan spools and one core spool, for example). The program, written in Fortran IV language, uses performance maps (in Block Data format) of the major engine components. Information on setting up the Block Data and input data is given in the report. Also included is a complete program listing with a description of each subroutine and sample results.

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SYMBOLS

STATION NUMBERS

| | |
|----|---|
| 1 | ambient |
| 2 | fan entrance |
| 21 | fan exit/intermediate compressor and duct entrance |
| 22 | intermediate compressor exit/high compressor entrance |
| 3 | high compressor exit/burner entrance |
| 4 | burner exit/hi pressure turbine entrance |
| 45 | hi press. turbine exit/int. press. turbine entrance |
| 5 | int. turbine exit/low press. turbine entrance |
| 55 | low press. turbine exit |
| 6 | afterburner entrance |
| 7 | afterburner exit |
| 8 | main nozzle throat |
| 9 | main nozzle exit |
| 23 | duct burner entrance |
| 24 | duct burner exit |
| 25 | duct exit if mixed flow engine |
| 28 | duct nozzle throat |
| 29 | duct nozzle exit |

OTHER SYMBOLS

| | |
|---|-------------------------------|
| T | total temperature (degrees R) |
| P | total pressure (atm) |
| H | total enthalpy (Btu/lbm) |
| S | total entropy (Btu/lbm °R) |

| | |
|-----|--------------------------------|
| TS | static temperature (degrees R) |
| PS | static pressure (atm) |
| FAR | fuel-air ratio |
| AM | Mach number |
| V | velocity (ft/sec) |

COMPONENT SYMBOLS

| | |
|--------|----------------------------------|
| F | fan |
| I | intermediate (middle) compressor |
| C | high pressure compressor |
| B, COM | combustor |
| D, DUC | fan duct |
| THP | high pressure turbine |
| TIP | intermediate turbine |
| TLP | low pressure turbine |
| A, AFT | afterburner |
| NOZ | nozzle |
| M | main nozzle |
| OB | overboard |
| T | total |

ENGINE SYMBOLS

| | |
|------|--|
| ETAR | ram recovery, P_2/P_1 |
| CN | ratio of corrected speed to design corrected speed |
| PCN | percent of design shaft speed |
| PR | pressure ratio |

In addition to all input symbols, the following are output:

| | |
|-------|---|
| A | areas (ft ²) |
| AM | Mach numbers |
| BLDU | total bleed flow into duct (lbm/sec) |
| BLDUC | bleed flow into duct from high-pressure compressor (lbm/sec) |
| BLDUI | bleed flow into duct from intermediate compressor (lbm/sec) |
| BLC | high-pressure compressor bleed flow (lbm/sec) |
| BLF | fan bleed flow, overboard (lbm/sec) |
| BLHP | total bleed flow to high-pressure turbine (lbm/sec) |
| BLHPC | high-pressure turbine cooling bleed from high-pressure compressor (lbm/sec) |
| BLHPI | high-pressure turbine cooling bleed from intermediate compressor (lbm/sec) |
| BLI | intermediate compressor bleed flow (lbm/sec) |
| BLIP | total bleed flow to intermediate turbine (lbm/sec) |
| BLIPC | intermediate turbine cooling bleed from high-pressure compressor (lbm/sec) |
| BLIPI | intermediate turbine cooling bleed from intermediate compressor (lbm/sec) |
| BLLP | total bleed flow to low-pressure turbine (lbm/sec) |
| BLLPC | low-pressure turbine cooling bleed from high-pressure compressor (lbm/sec) |
| BLLPI | low-pressure turbine cooling bleed from intermediate compressor (lbm/sec) |
| BLOB | total overboard bleed flow (lbm/sec) |
| BLOBC | overboard bleed flow from high-pressure compressor (lbm/sec) |
| BLOBI | overboard bleed flow from intermediate compressor (lbm/sec) |

| | |
|--------|---|
| FGP | pressure thrust (lbf) |
| FN | net thrust (lbf) |
| FRD | ram drag (lbf) |
| GU | initial or guessed values |
| ITRYS | number of loops thru engine before quitting |
| LOOPER | number of loops thru engine |
| SFC | specific fuel consumption (lbm/lbf/hr) |
| TOLALL | tolerance on convergence |
| VA | flight speed (ft/sec) |
| VJ | jet velocity (ft/sec) |

INPUT SYMBOLS

| | |
|-------------------------|--|
| ALTP | altitude (ft) |
| AM | flight Mach number |
| AM55 | Mach number at low pressure turbine exit |
| AM6 | afterburner entrance Mach number at design |
| A8 | main nozzle throat area - can be changed at off design - (ft ²) |
| A28 | duct nozzle throat area - can be changed at off design - (ft ²) |
| CNHPDS | design corrected speed - high pressure turbine |
| CNIPDS | design corrected speed - intermediate turbine |
| CNLPDS | design corrected speed - low pressure turbine |
| CVDNOZ | duct nozzle velocity coefficient |
| CVMNOZ | main nozzle velocity coefficient |
| DELFG, DELFN, DELSFC | delta degradation multiplier for gross thrust, net thrust, and specific fuel consumption, respectively. Usually input as 1.0 |
| DPAFDS | afterburner design pressure drop, $\Delta P/P$ |
| DPCODS | combustor design pressure drop, $\Delta P/P$ |

| | |
|---------|---|
| DTCODS | combustor design temperature rise, ΔT ($^{\circ}R$) |
| ETAA | afterburner efficiency |
| ETABDS | combustor efficiency at design |
| ETACDS | high-pressure compressor adiabatic efficiency at design |
| ETAD | duct burner efficiency |
| ETAFDS | fan adiabatic efficiency at design |
| ETAIDS | intermediate compressor adiabatic efficiency at design |
| ETAR | ram recovery, $P2/P1$ |
| ETHPDS | high-pressure turbine adiabatic efficiency at design |
| ETIPDS | intermediate turbine adiabatic efficiency at design |
| ETLPDS | low-pressure turbine adiabatic efficiency at design |
| HPEXT | horsepower extraction |
| LAFTBN | index on afterburning desired |
| LAMTP | index on ram or inlet operation desired |
| IDBURN | index on duct burning desired |
| IDCD | duct nozzle will be convergent-divergent when $IDCD=1$ |
| IDES | index for design point; must be set equal to 1 to design engine; zeroed automatically |
| IDUMP | index for dumping error matrix |
| IGASM | index for mixed flow or non-mixed flow turbofans |
| IMCD | main nozzle will be convergent-divergent when $IMCD=1$ |
| ITRYS | index for maximum number of iterations |
| MODE | independent variable designator for engine operation |
| NOZFLT | index for floating main or duct nozzle |
| PCBLC | % of total high-pressure compressor airflow that is bled |
| PCBLDUC | % of high-pressure compressor bleed flow that is bled into duct |

| | |
|---------|--|
| PCBLDUI | % of intermediate compressor bleed flow that is bled into duct |
| PCBLF | % of total fan airflow that is bled (overboard) |
| PCBLHPC | % of high-pressure compressor bleed flow that is bled to the high-pressure turbine |
| PCBLHPI | % of intermediate compressor bleed flow that is bled to the high pressure turbine |
| PCBLI | % of total intermediate compressor airflow that is bled |
| PCBLIPC | % of high-pressure compressor bleed flow that is bled to the intermediate turbine |
| PCBLIPI | % of intermediate compressor bleed flow that is bled to the intermediate turbine |
| PCBLLPC | % of high-pressure compressor bleed flow that is bled to the low-pressure turbine |
| PCBLLPI | % of intermediate compressor bleed flow that is bled to the low-pressure turbine |
| PCBLOBC | % of high pressure compressor bleed flow that is bled overboard |
| PCBLOBI | % of intermediate compressor bleed flow that is bled overboard |
| PCNC | high-pressure compressor shaft speed as a percent |
| PCNCDS | design high-pressure compressor shaft speed as a percent |
| PCNF | fan shaft speed as a percent |
| PCNFDS | design fan shaft speed as a percent |
| PCNI | intermediate compressor shaft speed as a percent |
| PCNIDS | design intermediate compressor shaft speed as a percent |
| PRCDS | high-pressure compressor pressure ratio at design |
| PRFDS | fan pressure ratio at design |
| PRIDS | intermediate compressor pressure ratio at design |
| PS55 | static pressure at low-pressure turbine exit (atm) |
| P2 | fan face total pressure, for nonstandard days (atm) |

| | |
|---------------------|---|
| TFHPDS | high-pressure turbine flow function at design |
| TFIPDS | intermediate turbine flow function at design |
| TFLPDS | low-pressure turbine flow function at design |
| TOLALL | tolerance on convergence at error matrix |
| T2 | fan face total temperature, for nonstandard days (°R) |
| T24 | duct-burner exit temperature, when ductburning (°R) |
| T4 | combustor exit/turbine inlet temperature (°R) |
| T4DS | combustor exit/turbine inlet temperature at design (°R) |
| T7 | afterburner exit temperature, when afterburning (°R) |
| WFA | afterburner fuel flow rate, IAFBN=2 (lbm/sec) |
| WFB | main combustor fuel flow rate, MODE=2 (lbm/sec) |
| WFBDS | combustor fuel flow rate at design, MODE=2 (lbm/sec) |
| WFD | duct burner fuel flow rate, IDBURN=2 (lbm/sec) |
| ZCDS, ZFDS, ZIDS | design ratio of high pressure compressor, fan, and intermediate compressor respectively; equals pressure ratio at design on design speed line minus one divided by high(surge) value minus one on the design speed line |

OUTPUT SYMBOLS¹

¹

Some symbols, such as T, are followed by station numbers; see station number symbols

In addition to all input symbols, the following are output:

| | |
|-------|---|
| A | areas (ft ²) |
| AM | Mach numbers |
| BLDU | total bleed flow into duct (lbm/sec) |
| BLDUC | bleed flow into duct from high-pressure compressor (lbm/sec) |
| BLDUI | bleed flow into duct from intermediate compressor (lbm/sec) |
| BLC | high-pressure compressor bleed flow (lbm/sec) |
| BLF | fan bleed flow, overboard (lbm/sec) |
| BLHP | total bleed flow to high-pressure turbine (lbm/sec) |
| BLHPC | high-pressure turbine cooling bleed from high-pressure compressor (lbm/sec) |
| BLHPI | high-pressure turbine cooling bleed from intermediate compressor (lbm/sec) |
| BLI | intermediate compressor bleed flow (lbm/sec) |
| BLIP | total bleed flow to intermediate turbine (lbm/sec) |
| BLIPC | intermediate turbine cooling bleed from high-pressure compressor (lbm/sec) |
| BLIPI | intermediate turbine cooling bleed from intermediate compressor (lbm/sec) |
| BLLP | total bleed flow to low-pressure turbine (lbm/sec) |
| BLLPC | low-pressure turbine cooling bleed from high-pressure compressor (lbm/sec) |
| BLLPI | low-pressure turbine cooling bleed from intermediate compressor (lbm/sec) |
| BLOB | total overboard bleed flow (lbm/sec) |
| BLOBC | overboard bleed flow from high-pressure compressor (lbm/sec) |
| BLOBI | overboard bleed flow from intermediate compressor (lbm/sec) |

| | |
|--------|--|
| BYPASS | bypass ratio, duct airflow divided by intermediate compressor airflow |
| CNC | high pressure compressor corrected shaft speed as a percent |
| CNF | fan corrected shaft speed as a percent |
| CNHP | high pressure turbine corrected shaft speed as a percent, $PCNC/\sqrt{T4}$ |
| CNHPCF | high pressure turbine speed correction factor |
| CNI | intermediate compressor corrected speed as a percent |
| CNIP | intermediate turbine corrected shaft speed as a percent, $PCNI/\sqrt{T45}$ |
| CNIPCF | intermediate turbine speed correction factor |
| CNL | low-pressure turbine corrected shaft speed as a percent, $PCNF/\sqrt{T5}$ |
| CNLPCF | low-pressure turbine speed correction factor |
| CS | ambient speed at sound (ft/sec) |
| DHHPCF | high-pressure turbine work correction factor |
| DHIPC | intermediate turbine work correction factor |
| DHLPCF | low-pressure turbine work correction factor |
| DHTC | high-pressure turbine work, ΔH , (Btu/lbm) |
| DHTCHP | high-pressure turbine work, temperature corrected, $\Delta H/T$ (Btu/lbm °R) |
| DHTCIP | intermediate turbine work, temperature corrected, $\Delta H/T$ (Btu/lbm °R) |
| DHTCLP | low-pressure turbine work, temperature corrected, $\Delta H/T$ (Btu/lbm °R) |
| DHTF | low-pressure turbine work, ΔH (Btu/lbm) |
| DHTI | intermediate turbine work, ΔH (Btu/lbm) |
| DPAFT | afterburner/tailpipe pressure, loss, $\Delta P/P$ |
| DPCOM | combustor pressure loss, $\Delta P/P$ |
| DPDUC | fan duct pressure loss, $\Delta P/P$ |

| | |
|---------|--|
| DTAFCF | afterburner temperature rise correction factor |
| DTAFDS | afterburner temperature rise at design, ΔT ($^{\circ}R$) |
| DTCOCF | Combustor temperature rise correction factor |
| DTDUCF | duct-burner temperature rise correction factor |
| DTDUDS | duct-burner temperature rise at design, ΔT ($^{\circ}R$) |
| ETAACF | afterburner efficiency correction factor |
| ETAADS | afterburner efficiency at design |
| ETAB | combustor efficiency |
| ETABCF | combustor efficiency correction factor |
| ETAC | high-pressure compressor adiabatic efficiency |
| ETACCF | high-pressure compressor efficiency correction factor |
| ETAD | ductburner efficiency |
| ETADCF | duct burner efficiency correction factor |
| ETADDS | ductburner efficiency at design |
| ETAF | fan adiabatic efficiency |
| ETAFCF | fan efficiency correction factor |
| ETAI | intermediate compressor adiabatic efficiency |
| ETAI CF | intermediate compressor efficiency correction factor |
| ETATHP | high-pressure turbine adiabatic efficiency |
| ETATIP | intermediate turbine adiabatic efficiency |
| ETATLP | low-pressure turbine adiabatic efficiency |
| ETHPCF | high-pressure turbine efficiency correction factor |
| ETIPCF | intermediate turbine efficiency correction factor |
| ETLPCF | low-pressure turbine efficiency correction factor |
| FAR | fuel-air ratios |
| FART | total fuel-air ratio |
| FG | gross thrust (lbf) |

| | |
|--------|---|
| FGM | momentum thrust, total (lbf) |
| FGMD | momentum thrust from duct (lbf) |
| FGMM | momentum thrust from core (lbf) |
| FGP | total pressure thrust (lbf) |
| FGPD | pressure thrust from duct (lbf) |
| FGPM | pressure thrust from core (lbf) |
| FN | net thrust (lbf) |
| FRD | ram drag (lbf) |
| H | enthalpies (Btu/lbm) |
| P | total pressures (atm) |
| PCNC | high-pressure compressor shaft speed as a percent |
| PCNCGU | first guess of PCNC |
| PCNF | fan shaft speed as a percent |
| PCNFGU | first guess at PCNF |
| PCNI | intermediate compressor shaft speed as a percent |
| PCNIGU | first guess at PCNI |
| PRC | high pressure compressor pressure ratio |
| PRCCF | high pressure compressor pressure ratio correction factor |
| PRF | fan pressure ratio |
| PRFCT | fan pressure ratio correction factor |
| PRI | intermediate compressor pressure ratio |
| PRICF | intermediate compressor pressure ratio correction factor |
| PS | static pressures (atm) |
| S | entropies (Btu/lbm - °R) |
| SFC | specific fuel consumption (lbm/lbf/hr) |
| T | total temperatures (°R) |

| | |
|--------|--|
| TS | static temperatures |
| T2DS | fan face total temperature at design ($^{\circ}$ R) |
| T21DS | fan exit total temperature at design ($^{\circ}$ R) |
| T22DS | intermediate compressor exit total temperature at design ($^{\circ}$ R) |
| T24DS | ductburner exit temperature at design ($^{\circ}$ R) |
| T7DS | afterburner exit temperature at design ($^{\circ}$ R) |
| T4GU | first guess at T4 ($^{\circ}$ R) |
| TFFHP | high-pressure turbine flow function |
| TFFIP | intermediate turbine flow function |
| TFFLP | low-pressure turbine flow function |
| TFHPCF | high-pressure turbine flow function correction factor |
| TFIPCF | intermediate turbine flow function correction factor |
| TFLPCF | low-pressure turbine flow function correction factor |
| V | velocities (ft/sec) |
| VA | aircraft velocity (ft/sec) |
| VJD | fan duct exhaust velocity (ft/sec) |
| VJM | core exhaust velocity (ft/sec) |
| WA | airflows (lbm/sec) |
| WAC | high-pressure compressor airflow (lbm/sec) |
| WACC | high-pressure compressor corrected airflow (lbm/sec) |
| WACCF | high-pressure compressor airflow correction factor |
| WACDS | high-pressure compressor airflow at design (lbm/sec) |
| WAD | duct airflow (lbm/sec) |
| WAF | fan airflow (lbm/sec) |
| W AFC | fan corrected airflow (lbm/sec) |
| WAFCF | fan airflow correction factor |

| | |
|------------|---|
| WAI | intermediate compressor airflow (lbm/sec) |
| WAIC | intermediate compressor corrected airflow (lbm/sec) |
| WAICF | intermediate compressor airflow correction factor |
| WA3CDS | high-pressure compressor exit corrected airflow at design (lbm/sec) |
| WA23CDS | ductburner entrance corrected airflow at design (lbm/sec) |
| WFADS | afterburner fuel flow rate at design (lbm/sec) |
| WFDDS | ductburner fuel flow rate at design (lbm/sec) |
| WFT | total fuel flow rate (lbm/sec) |
| WGT | total gas flow rate (lbm/sec) |
| WG6CDS | afterburner entrance corrected gas flow at design (lbm/sec) |
| ZC, ZF, ZI | similar to ZCDS, ZFDS, and ZIDS except these are off-design - see Input Symbols |

SECTION I
INTRODUCTION

In 1967, a digital computer program for balancing cycle turbofan engines titled SMOTE was developed in the Air Force Aero Propulsion Laboratory (Reference 1). This program is capable of simulating design and off-design performance of two-spool turbofan engines. This report describes a similar technique, obtained by modifying the SMOTE program, that simulates design and off-design performance of three-spool turbofan engines. NASA, Pratt and Whitney, General Electric and other contractors have developed or are developing working models that simulate the performance of three spool engines. However, through the experience gained in developing an in-house computer program, more flexibility can be exercised in making modifications to satisfy varying future needs without being dependent on contractors or other organizations.

Three-spool turbofan engines will be competitive as candidates for powering future aircraft systems. This effort has provided the Air Force Aero Propulsion Laboratory with an in-house method for evaluating the performance of three-spool turbofan engines.

SECTION II

SUMMARY

A computer program titled TRISPL is described. This program is derived from SMOTE (Simulation of Turbofan Engine) which was developed by the Turbine Engine Division of the AF Aero Propulsion Laboratory, Wright-Patterson AFB, Ohio.

TRISPL calculates design and off-design performance of 3-spool turbofan engines. Component maps, input as block data, are scaled internally to simulate a specific engine. The program is formulated for two core spools and one fan spool operation. Options are included for mixed or separate flow engines and dry or afterburning operation. The program can be modified for different modes of operation and engine types, 2 fan spools and one core spool for example.

SECTION III

METHOD OF ENGINE CALCULATIONS

The following discussion is very similar to that in the report describing the SMOTE program.

1. COMPONENT MAPS

The performance of the major engine components is based on component maps. These maps are usually obtained from analytical methods or rig-testing and are then converted into Block Data subroutines for use by TRISPL. The maps presently included in TRISPL are very general and do not represent any particular engine or engine components.

The component maps are scaled at the engine design point by TRISPL in order to match their performance to a desired set of performance figures which are input as data. Scaling or correction factors are calculated and then applied to the maps at off-design points. The scaling process is linear; therefore, correction factors near unity result in the highest accuracy of component simulation. Conversely, however, not being close to 1.0 does not necessarily mean that the simulation is poor since many maps have been shown to be typical over quite large ranges in the variables.

TRISPL presently includes component maps for the fan, intermediate compressor, high pressure compressor, combustor, and the three turbines. Duct burning, duct losses, gas mixing, afterburning, tailpipe losses, and nozzle losses are all calculated or input, but these characteristics could also be included as Block Data if maps were available. Likewise, schedules for bleed air and variable area nozzles could be used.

a. Fan-Compressor Maps

The fan and compressor maps are very similar and are plots of pressure ratio vs. corrected airflow with constant corrected speed lines and constant efficiency islands (see Fig. 1). Entry to the map is through the corrected speed and Z , where Z is a ratio of pressure ratios, and is defined at a constant corrected speed as shown in Figure 1. It is advantageous to use Z instead of pressure ratio because Z is restrained between the limits of 0 and 1, whereas the limits on pressure ratio vary depending upon map location and the particular map. Also, as indication that the fan or a compressor is approaching surge is given as Z approaches 1.

b. Combustor Map

The combustor map is a plot of efficiency vs. temperature rise for constant input pressure (see Fig. 2). Entry to the map is through temperature rise and input pressure, with efficiency being output.

c. Turbine Maps

The turbine map is a plot of work function vs. corrected speed with constant turbine flow function lines and constant efficiency islands (see Fig. 3). The work function and flow function are defined as

$$\text{DHTC} = \frac{H_{\text{IN}} - H_{\text{OUT}}}{T_{\text{IN}}}$$

and

$$\frac{WG_{\text{IN}} \sqrt{T_{\text{IN}}}}{P_{\text{IN}}}$$

Entry to the map is through corrected speed and flow function, with work function and efficiency being output.

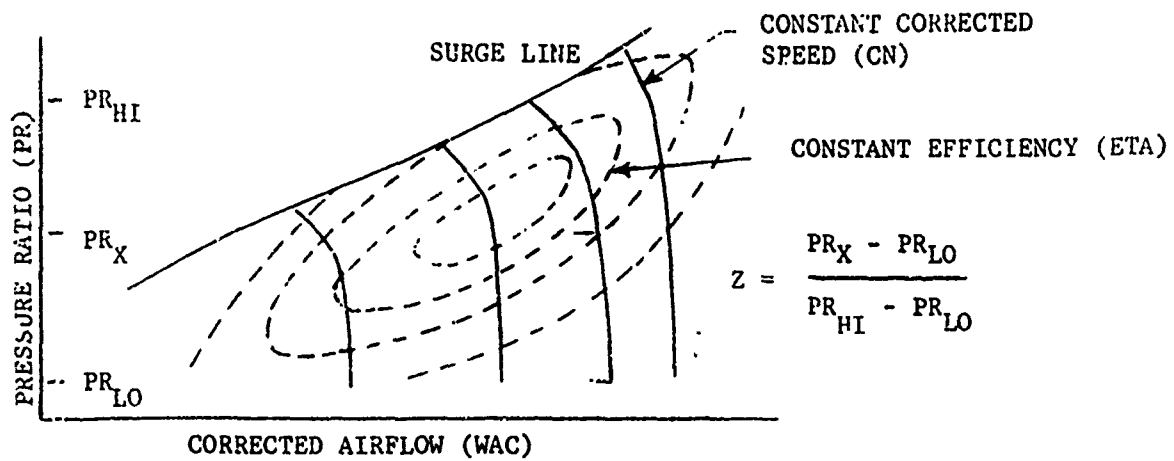


Figure 1. Example of Fan-Compressor Map

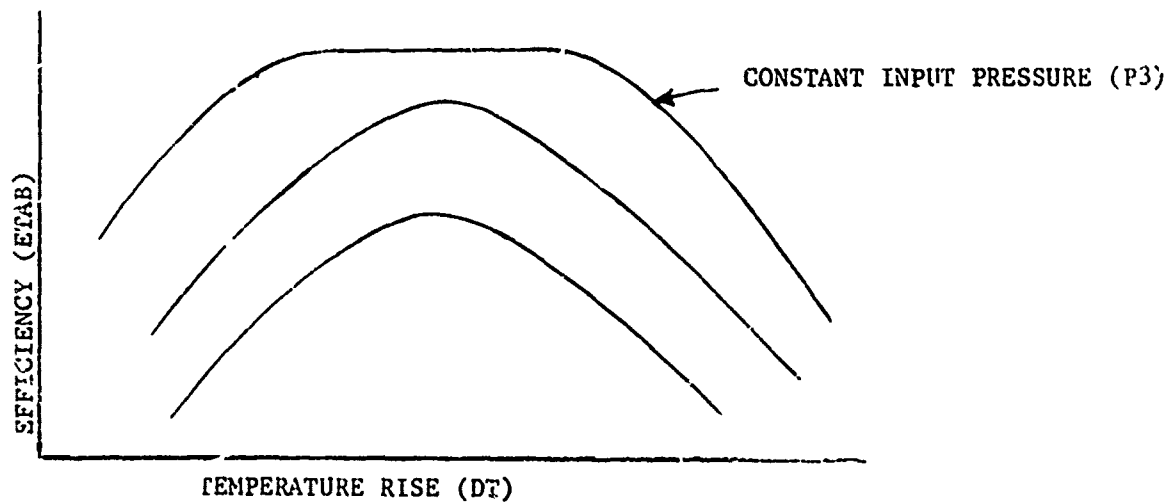


Figure 2. Example of Combustor Map

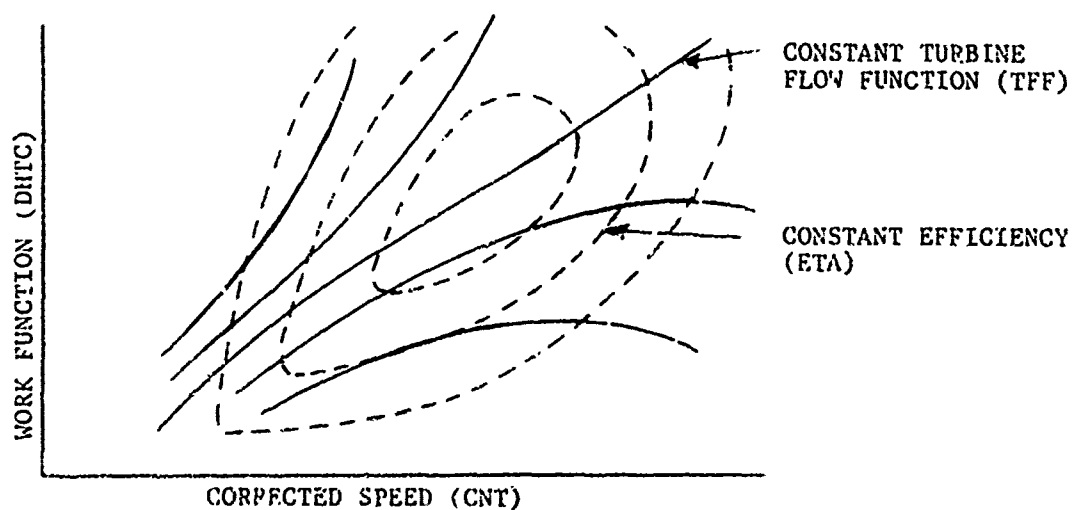


Figure 3. Example of Turbine Map

The work function could have been used as an entry in place of one of the present entries, but, because of the shape of the curves, this could lead to double entry points for one work function. However, if the turbine maps were plotted in a different format, this could be an acceptable method.

2. DESIGN POINT

Once the component maps have been reduced to Block Data form and placed in the program, it is necessary to run a design point. The design point is run at those conditions under which the real engine is designed or sized, often sea level static. Design parameters necessary to simulate the real engine (for example, airflow, bypass ratio, turbine inlet temperature, various pressure losses, pressure ratios, etc.) are input and a complete thermodynamic cycle calculation is performed. For more details on the cycle calculation see Section III 4, "Off-Design Points". Scale factors for the component maps are calculated to insure that the input design parameters are met. If the design parameters have been correctly input, the design point will be completed after one pass through the engine calculations (that is, no balancing will occur) because the maps are shifted to reduce the errors to zero.

Other parameters calculated and output at the design point include certain temperatures, airflows, gas mixing areas, and nozzle throat and exit areas.

3. SCALING FACTORS

Scaling or correction factors are calculated at the design point using the following equation:

$$P (\text{correction factor}) = P (\text{design})/P (\text{map})$$

where P represents a general parameter. One exception to this equation

is the equation for calculating the fan and compressors pressure correction factors:

$$PR \text{ (correction factor)} = \frac{[PR \text{ (design)} - 1]}{[PR \text{ (map)} - 1]}$$

where PR represents a general pressure ratio.

Theoretically, if the component maps and the input design parameters are exact representations of a particular engine, the correction factors will equal 1. However, this will not be true due to map interpolations, certain assumptions such as ideal and isentropic flow, and tolerances in the thermodynamic calculations. If unmatched component maps are used, the correction factors can differ significantly from 1.

4. OFF-DESIGN POINTS

The following discussion pertains particularly to off-design points, although the input and the general cycle calculations are the same for the design point. Throughout the following discussion, it should be remembered that scaling or correction factors (multipliers) are applied to all performance maps (Block Data parameters). A schematic diagram of the engine components and station designations is shown in Fig. 4.

a. Input

The program uses a controlled output; that is, the variables desired as output can be selected at the start of a run. This selection is obtained by placing the names of the variables in the first section of input cards. Controls, scaling factors and operating conditions make up the rest of the input.

The control inputs are used to determine the type of engine; mixed or separate flow, afterburning or duct burning, and convergent or convergent-divergent nozzle. The controls are also used to fix the

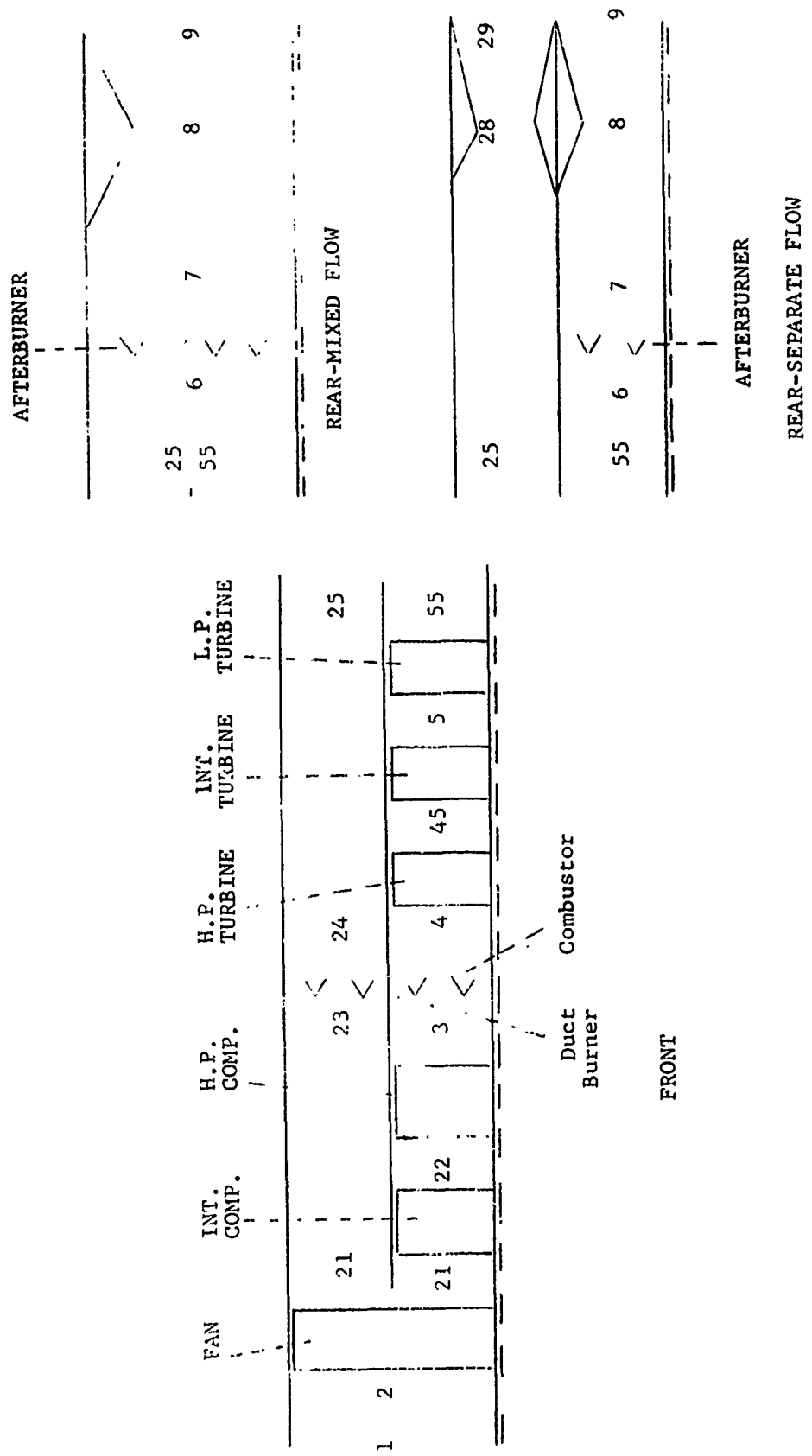


Figure 4. Schematic of Engine Components

mode of operation; constant PCNF, constant PCNC, constant T4, or constant WFB. Other controls determine inlet conditions, title printout, and cycle looping printouts. The correction factors can be input directly, or the design point can be run first and the calculated factors will be stored in common. The operating conditions include the flight Mach number, altitude, power setting (either PCNF, PCNC, T4, or WFB), duct burner and afterburner temperatures or fuel flows, bleed, and horsepower extraction.

b. Initial Values

The program uses six primary independent variables: ZF, PCNF, ZI, PCNI, ZC, PCNC (T4 may be substituted for PCNF or PCNC, depending upon the mode of operation). Three secondary independent variables (TFFHP, TFFIP, and TFFLP) are also used to ensure correct entry into the turbine maps. Initial values for these nine variables must be obtained to start the program at each point. A subroutine (GUESS) supplies these variables as a function of T2, T21, and some of the variables themselves. It is important to note that the closer the initial values are to the final values at a balanced point, the faster the program will run. Therefore, after a particular engine configuration has been run a few times, it is usually advisable to change the general initial value equations to suit the engine, using the knowledge gained from past runs to estimate more closely the final values of the variables.

c. Inlet

The thermodynamic properties of the atmosphere are found from a 1962 ARDC Atmosphere Tables subroutine. Using conservation of energy and isentropic flow, the conditions at the face of the fan can be found. A ram recovery can be input or, if not input, a ram recovery defined by MIL-E-5008E Specifications will be used. If desired, a T2-P2 direct input mode is available as are provisions for nonstandard day conditions.

d. Fan and Compressor

Block Data is used to determine the performance characteristics of the fan and compressors. When Z and PCN are known, the pressure ratio, corrected airflow and efficiency can be found by using a general Block Data interpolation routine named SEARCH. With the pressure ratio known and when the assumption of isentropic compression and the efficiency are used, the thermodynamic conditions at the exit of the fan and compressors can be calculated. Bleed for consumer use, leakage, or cooling is accounted for. Actual airflow leaving the fan and compressors is calculated from corrected airflow, temperature, pressure, and bleed.

It should be mentioned here that the present form of TRISPL calculates corrected speed (CN) in a manner slightly different from SMOTE. TRISPL uses an equation which forces corrected speed (CN) to equal physical shaft speed (PCN) at the design point. The equation is

$$CN = (PCN/\sqrt{\theta}) \cdot \sqrt{\theta_D}$$

where θ is corrected temperature, and θ_D is the corrected temperature at design. Thus, at the design point, $CN = PCN$. This equation is useful for studying theoretical engines where actual maps are not available but must be changed back to the SMOTE equation if real engines are to be simulated. SMOTE uses this equation: $CN = PCN/\sqrt{\theta}$.

e. Combustor

The pressure drop in the combustor is a function of a design pressure drop and ratio of corrected airflow to design corrected airflow.

Combustor efficiency is obtained from Block Data using SEARCH. The fuel used is assumed to be JP-4 (at 59°F), and, with the assumption of adiabatic and constant pressure combustion, a fuel heating value equation as a function of T4 has been derived. Thus the fuel/air ratio, fuel flow, and thermodynamic conditions at the combustor exit can be calculated. If WFB is known instead of T4, a small iteration is necessary.

f. Turbines

The turbine subroutines all use similar logic and obtain their performance characteristics from Block Data using subroutine SEARCH. All three turbine parameters (CN, TFF, DHTC) can be calculated before entering the turbine map, but only two are needed. Therefore, the third parameter obtained from the map is compared with the calculated third parameter, and a balancing error is generated if they are not equal. In this program, CN and TFF are used for map entries, and DHTC is used to generate the error. In addition, the efficiency is also obtained through SEARCH.

In addition, another error will be generated if TFF is not within map limits. The error will be the difference between TFF and the nearest map limit. This error becomes particularly important when the estimated initial values of the independent variables are far from the correct values, and the point is extremely unbalanced. When either TFF or CN is not within map limits, they are set to the nearest map limit, and one of the independent variables is changed in an attempt to rectify the situation. The operating point must appear on all maps before a complete cycle calculation can be accomplished.

Horsepower extraction is accounted for in calculating DHTC of the high pressure turbine. When the efficiency is used and the turbine process is assumed isentropic, the thermodynamic properties at the three turbine exits can be calculated. Any bleed airflow for cooling the turbines is treated as if it entered the main stream behind the turbine, and the thermodynamic properties at the turbine exit are recalculated to account for this.

g. Duct

The duct airflow and bypass ratio are calculated from the fan and intermediate compressor airflows. The pressure drop in the duct is treated as in the main combustor. For duct burning, the same fuel heating value equation that was used in the main combustor is again used, but the efficiency must be input. As in the combustor, either the temperature (T24) or the fuel flow (WFD) may be input.

If a separate flow engine is being simulated, the duct nozzle calculations are done in this routine, although they are accomplished in the same manner as for the main nozzle.

h. Mixer

The gas mixing areas (duct exit and turbine discharge for a mixed flow engine or just the turbine discharge area for a separate flow engine) are calculated at the design point using either an input core static pressure or Mach number. In the mixed flow mode, there is an option for calculating afterburner entrance area as a function of an input afterburner entrance Mach number at the design point. At an off-design point the areas are used to calculate static pressures and Mach numbers.

For a separate flow engine, the thermodynamic conditions entering the afterburner are now known, since they are identical to turbine discharge conditions.

For a mixed flow engine, a set of derived equations based on one-dimensional fluid flow theory and conservation of mass, energy, and momentum is used to determine the thermodynamic conditions after complete mixing of the two gas streams. These equations do not require that the static pressures of the two entering streams be equal. However, for a correct engine balance, the two static pressures must be equal, and a balancing error is generated if they are not equal.

i. Afterburner

The dry loss (cold loss) pressure drop in the afterburner is a function of a design pressure drop and the ratio of corrected gas flow to the design corrected gas flow.

For afterburning, the same equation for the fuel heating value that was used in the combustor is again used, but the efficiency must be input. As in the combustor, either temperature (T_2) or the fuel flow (WFA) may be input. A momentum loss (hot loss) pressure drop is also calculated.

j. Nozzle

The main nozzle program uses fixed effective areas (except when afterburning or when different nozzle areas are directly input) calculated at the design point. Either a convergent or convergent-divergent subroutine may be used depending upon the input controls. If afterburning has been selected, the nozzle areas are allowed to float to obtain optimum performance; however, the areas are returned to their original design values after the afterburning point is completed. Nozzle

areas can also be changed by directly inputting different nozzle area values. The duct nozzle behaves identically to the main nozzle, including floating areas if duct-burning has been selected.

Because all thermodynamic properties of the gas stream are known, as well as the amount of flow, nozzle areas, and ambient pressure, there is a redundant parameter. For this program, the total pressure of the gas stream was chosen as the redundant parameter. The nozzle calculations are made without using the total pressure, and a required total pressure compatible with all other known parameters is calculated. This required pressure is compared with the actual pressure and a balancing error is generated if they are not equal.

k. Performance and Output

At this point, nine errors have been generated after one pass through the engine. Several more passes must be completed under control of the error matrix and engine balancing technique. See Section IV for a detailed description of the balancing technique. Eventually however, the errors will be reduced to zero, and engine performance will be calculated using standard equations. Gross thrust is obtained by summing the momentum term (a nozzle velocity coefficient may be input) and a pressure-area term, and the net thrust is in turn found by subtracting a ram drag (airflow momentum loss at inlet) term from the gross thrust. Specific fuel consumption (SFC) is total fuel flow divided by net thrust.

As previously mentioned, a controlled output is used, whereby only selected variables are printed. Each variable is labeled with its name and provisions have been made for changing the name of a variable. In addition, the values of all variables in common are printed in a close format so that variables other than those selected for a specific run are available later on.

5. QUADRATIC INTERPOLATION ROUTINE

Throughout the program there are many small loops (for example, thermodynamic iterations and table look-up) which require convergence. Trial-and-error methods and linear interpolations can be time consuming, especially when a tight tolerance is necessary; therefore, a general interpolation routine called AFQUIR (Air Force Quadratic Interpolation Routine) (Reference 1) is used.

This routine requires a dummy array dimensioned for nine locations. Also input into the routine through the calling argument are the independent and the dependent variables, the answer or value which the dependent variable is to converge upon, the number of tries at convergence, the tolerance, and a variable called DIR.

The DIR is either set or calculated in the calling program and is an initial guess at the direction and percentage change to apply to the first value of the independent variable. If not enough is known about the variable to calculate DIR, an arbitrary value may be set. This should not affect the final result, but may increase the number of tries of convergence.

The DIR thus establishes the second value at the independent variable. This value is used in the calling program to determine a corresponding second value of the dependent variable and AFQUIR is called a second time with two sets of values. A linear interpolation is made which results in a third value of the independent variable. AFQUIR is then called a third time with the third values of the independent and dependent variables and a quadratic interpolation is made. The values of these three sets of variables have been stored in the dummy array, and from hereon, quadratic

interpolations are made using the three sets which give values closest to the answer. Values farthest from the answer are lost.

Various safeguards are built into AFQUIR to return the interpolation method to DIR or linear if the roots of the quadratic become complex, if the quadratic does not intercept the answer, if the value of the independent variable differs radically from previous values, or if two sets of independent and dependent variables are identical.

Also it is possible to preload the dummy array directly at the linear or quadratic interpolations if desired.

In summary, AFQUIR is a completely flexible routine which performs quadratic interpolation for quick convergence of general functions.

SECTION IV
BALANCING TECHNIQUE

The balancing technique is virtually the same as that used in SMOTE. It is based on finding a solution for a set of partial differential equations. For this program, the set is composed of nine equations; however, using a set of only three equations will simplify the following discussion. This corresponds to a basic single spool turbojet simulation. SMOTE uses a set of six equations.

As discussed previously, nine independent variables (ZF, PCNF or T4, ZI, PCNI, ZC, PCNC or T4, TFFHP, TFFIP, and TFFLP) were selected. Once these variables have been given initial values, it is possible to proceed through an entire engine cycle calculation. Nine errors are generated as shown in Section III. These initial values of the nine variables and nine errors are referred to as base values.

In the following equations, V refers to a variable and E to an error. The basic set of differential equations based on $E = f(V)$ is

$$dE_1 = \frac{\partial E_{11}}{\partial V_1} dV_1 + \frac{\partial E_{12}}{\partial V_2} dV_2 + \frac{\partial E_{13}}{\partial V_3} dV_3$$

$$dE_2 = \frac{\partial E_{21}}{\partial V_1} dV_1 + \frac{\partial E_{22}}{\partial V_2} dV_2 + \frac{\partial E_{23}}{\partial V_3} dV_3$$

$$dE_3 = \frac{\partial E_{31}}{\partial V_1} dV_1 + \frac{\partial E_{32}}{\partial V_2} dV_2 + \frac{\partial E_{33}}{\partial V_3} dV_3$$

where the single subscripts correspond to three variables and three errors and where the double subscripts indicate the change in a particular error (first subscript) due to a change in a particular variable (second subscript).

Assuming small changes result in the following approximations (where B refers to a base value):

$$dE = E - EB$$

$$dV = V - VB$$

$$\frac{\partial E}{\partial V} = \frac{\Delta E}{\Delta V}$$

With these approximations and the fact that E should be zero when the engine is balanced, the set of partial differential equations reduces to

$$E_1 - EB_1 = \frac{\Delta E_{11}}{\Delta V_1} dV_1 + \frac{\Delta E_{12}}{\Delta V_2} dV_2 + \frac{\Delta E_{13}}{\Delta V_3} dV_3 = -EB_1$$

$$E_2 - EB_2 = \frac{\Delta E_{21}}{\Delta V_1} dV_1 + \frac{\Delta E_{22}}{\Delta V_2} dV_2 + \frac{\Delta E_{23}}{\Delta V_3} dV_3 = -EB_2$$

$$E_3 - EB_3 = \frac{\Delta E_{31}}{\Delta V_1} dV_1 + \frac{\Delta E_{32}}{\Delta V_2} dV_2 + \frac{\Delta E_{33}}{\Delta V_3} dV_3 = -EB_3$$

Three more passes (nine for TRISPL) are now made through the engine cycle calculations, and one variable is changed by a small amount (ΔV) for each pass. The change in each error due to the small change in the variables ($\Delta E/\Delta V$) can be calculated.

The above set of differential equations can now be solved for dV_1 , dV_2 and dV_3 and, in general, the new value of each independent variable would be given by

$$V = VB + dV$$

If the engine cycle calculations were linear functions, the engine would balance (errors equal zero) with these new values of the variables. However, this is usually not the case. The new errors become base errors (still keeping the old $\Delta E/\Delta V$'s) and another attempt at balance is performed. If several such attempts still fail, the entire process is repeated where the new errors and variables become base values and a new set of $\Delta E/\Delta V$'s are calculated.

A subroutine to determine the solution of a matrix is used to solve the set of differential equations. After each pass through the engine, a matrix array is loaded with the appropriate values; after ten passes (base value plus nine independent variables) the matrix subroutine is called to solve the matrix.

It was found that the "dV's" obtained from the solution of the differential equations were in many cases too large, thus causing the variables to exceed their limits, and to make it practically impossible to balance the cycle. The "dV's" are therefore multiplied by a suppression factor calculated in the program which limits the swing of the variables. Although this procedure may tend to increase the number of passes before balancing in some cases, it also balances points which previously would not balance. These points are most generally far from the design point, where oscillations of the dependent variables tend to build up.

SECTION V
INPUT/OUTPUT DESCRIPTIONS

1. BLOCK DATA INPUT

The three compressor maps are entered into the program as BLOCK DATA subprograms FANDAT, INTDAT, and CMPDAT.

Using FANDAT as an example (refer to program listing), and referring to typical map (Fig. 1), the data are programmed as follows:

Card 1 identifies the program as BLOCK DATA. Card 2 is a comment card. Card 3 identifies the common block FAN into which data are to be stored and dimensions the program variables. Card 4 indicates that there are 10 speed lines N and the number of points NP on each line (6 on the lowest speed, 7 on the next 3 lines, etc). Card 5 assigns the value of speed to each of the 10 lines (low to high). The remaining cards indicate the values of pressure ratio (PR), corrected airflow (WAC), and efficiency (ETA) for the speed lines. For example, the card

DATA (PR (4,J), J = 1,7)/

denotes that the pressure ratios are for the 4th speed line (CN = 0.6) and that there are 7 points.

The combustor BLOCK DATA subprogram is CMBDAT. Referring to the program listing and a typical combustor map (Fig 2), the data are programmed as follows:

Card 1 identifies the program as BLOCK DATA. Card 2 identifies the common block COMB into which data are to be stored and dimensions the variables. Card 3 indicates that there are 15 lines of constant PSI(P3) by the value of N, and that there are 15 values of DELT (DT) and ETA(ETAB) along

each line of constant PSI(P3). Cards 4 and 5 assign values to each of the P3 lines from low to high pressure. Cards 6 to 8 assign values of ΔT to each of the P3 lines starting at low ΔT . The lowest value of ΔT on each of ΔT on the lowest value of P3. Next comes the second lowest value of ΔT on each P3, etc. Cards 9 to 16 assign the value of ETAB in a one-to-one correspondence with the ΔT values just assigned. The order is the same.

The turbine maps are the BLOCK DATA subprograms HTURB, ITURB, and LTURB. Taking HTURB as an example and referring to the program listing and a typical map (Fig.3) the data are programmed as follows: Card 1 identifies the subprogram as BLOCK DATA. Card 2 is a COMMENT card. Card 3 identifies the common block HTURB into which data are to be loaded and dimensions the program variables. Card 4 indicates the number of constant turbine flow function lines TFF as 11 (N) and the number of points on each line from low to high TFF. Cards 5 and 6 set values of TFF from low to high. The remaining cards set the values of corrected speed (CN), work function (DH), and efficiency (ETA) starting from low TFF. For example, the card,

```
DATA (DH(5,J), J = 1,15)/
```

denotes that the work functions are for the 5th flow function line (TFF = 49.175) and that there are 15 points.

2. CONTROLLED OUTPUT/NAMELIST INPUT

The input data is divided into two sections; data cards for the controlled output, and data cards in Namelist format for running each point. For the following discussion on setting up input data, refer to the listing of sample data immediately following the program listing.

a. Controlled Output

The variables that are to be output are selected by the first section of data cards. Any variable that is in one of the main commons (DESIGN, FRONT, SIDE, or BACK) may be selected for output by punching the name at the variable as it appears in the common (with trailing blanks if necessary) in Columns 1 through 7. Up to 150 variables (25 lines of 6 variables) may be chosen for a particular run. During the output phase, the name of the variable is printed out, with its value printed immediately below the name.

Another feature of the controlled output is the ability to change the name of a variable to be output; for example, it may be desired to change a station designation to one more common to a particular programmer. In this case, the variable name would be punched in Columns 1 through 7 as described above, but in addition, the desired name would be punched in Columns 15 through 22. Special symbols, such as /, may be used in the new name. The last card of the controlled output must be a card with THEEND punched in Columns 1 through 6.

In addition to the variables selected as controlled output, the values of all variables in common are printed in a close format so that variables other than those selected for a specific run are available later on.

b. NAMELIST Input

The normal data for running the desired points follows the controlled output data and is in a Namelist format, where the name of the Namelist is DATAIN. Usually the first set of data is the design point, as shown in the sample input data. When the design point is run (DDES = 1), all map scaling or correction factors are printed out, as well as being retained

in common. Therefore, it is possible to run off-design points immediately following the design point by making use of the values in common, or to begin running an off-design point immediately by inputting the scaling or correction factors. The first method is usually easier, but the second method may be desired if many points are to be run using the same engine parameters with no changes except for power setting, Mach number, and altitude.

The variables that must be input at the design point for the basic cycle (for example, no afterburning) are listed in Table I below:

TABLE I
INPUTS REQUIRED FOR BASIC CYCLE AT DESIGN POINT

| VARIABLE | DEFINITION | UNITS |
|----------|---|------------------------------|
| PRFDS | Fan pressure ratio | |
| WAFDS | Fan face airflow | lb/sec |
| ETAFDS | Fan efficiency | |
| ZFDS | Design Z of fan | |
| PCNFDS | Fan shaft speed expressed as percent | |
| PRIDS | Intermediate compressor pressure ratio | |
| WAIDS | Intermediate compressor (core) airflow | lb/sec |
| ETAIDS | Intermediate compressor efficiency | |
| ZIDS | Design Z of intermediate compressor | |
| PCNIDS | Intermediate compressor shaft speed as a percent | |
| PRCDS | High pressure compressor pressure ratio | |
| ETACDS | High pressure compressor efficiency | |
| ZCDS | Design Z of high pressure compressor | |
| PCNCDS | High pressure compressor shaft speed as a percent | |
| ETABDS | Combustor efficiency | |
| DPCODS | Combustor pressure drop, $\Delta P/P$ | |
| DTCODS | Combustor temperature rise | $^{\circ}R$ |
| T4DS | Turbine inlet temperature | $^{\circ}R$ |
| TFHPDS | High pressure turbine flow function | $lb\sqrt{\sigma R}/(SFC)(P)$ |
| CNHPDS | High pressure turbine corrected speed | |
| ETHPDS | High pressure turbine efficiency | |
| TFIPDS | Intermediate turbine flow function | $lb\sqrt{\sigma R}/(SFC)(P)$ |

| <u>VARIABLE</u> | <u>DEFINITION</u> | <u>UNITS</u> |
|--------------------|--|---------------------------------|
| CNIPDS | Intermediate turbine corrected speed | |
| ETIPDS | Intermediate turbine efficiency | |
| TFLPDS | Low pressure turbine flow function | $lb\sqrt{\sigma R}/(sec)(psia)$ |
| CNLPDS | Low pressure turbine corrected speed | |
| ETLPDS | Low pressure turbine efficiency | |
| DPDUDS | Fan duct pressure drop, $\Delta P/P$ | |
| DPAFDS | Tailpipe pressure drop, $\Delta P/P$ | |
| AM55 or PS55 | Mach number at low pressure turbine exit Static pressure at low pressure turbine exit | atm |
| AM | Flight Mach number | |
| ALTP | Altitude | ft |
| HPEXT | Horsepower extraction | hp |
| CVMNOZ | Main nozzle velocity coefficient | |
| CVDNOZ | Duct nozzle velocity coefficient | |
| | + | |
| | Various bleed flows (see Symbols) | |
| | + | |
| | Various control parameters (see below) | |

As mentioned in Table I, various control parameters which fix the engine type, mode of operation, method of calculating ram recovery, etc. must be input. These are listed below. Subroutine ZERO determines what values in common will be zeroed between points. None of the design values or correction factors are zeroed but some of the control parameters are. In the control parameter listing below, the superscripts (1) to (4) have the following meanings: (i) automatically returned to zero after

each point is calculated, must be re-input if option is again desired, (2) option can be used only at off-design points; (3) these input values remain as input unless changed by a new value; (4) a set-up case must be run where all the components are first matched before these $\neq 0$ options are used, then the identical case may be repeated exercising these options.

| | | |
|-------------|-----|--|
| | (1) | |
| IDES = 1 | | For calculating design point |
| | (3) | |
| MODE = 0 | | Specify T4 |
| (2) | (3) | |
| MODE = 1 | | Specify PCNC |
| | (3) | |
| MODE = 2 | | Specify WFB |
| | (3) | |
| (2)MODE = 3 | | Specify PCNF |
| | (3) | |
| INIT = 0 | | Initializes point |
| | (3) | |
| INIT = 1 | | Will not initialize point |
| | (3) | |
| IDUMP = 0 | | No looping write-outs |
| | (3) | |
| IDUMP = 1 | | Will dump looping write-outs if error occurs |
| | (3) | |
| IDUMP = 2 | | Will dump looping write-outs after every point |
| | (3) | |
| IAMTP = 0 | | Will use AM and mil-spec ETAR |
| | (3) | |
| IAMTP = 1 | | Will use input AM and input ETAR |
| | (3) | |
| IAMTP = 2 | | Will use input T2 as T1=T1+T2 and standard P1 (T2 value needs to be input at every point or an error will occur whenever used) |
| | (3) | |
| IAMTP = 3 | | Will use input P2 and standard T1 |
| | (3) | |
| IAMTP=4 | | Will use input T2 and input P2 |
| | (3) | |
| IGASM=0 | | Separate flow, A6=A55 |

| | | |
|-------------------------|---------|--|
| IGASMX=1 | (3) | Will mix duct and main streams, $A6=A25+A55$ |
| IGASMX=2 | (3) | Will mix duct and main streams, input AM6 |
| IDBURN=1 | (4) (1) | For duct burning, input T24 |
| IDBURN=2 | (4) (1) | For duct burning input WFD |
| IAFTBN=1 | (4) (1) | For afterburning (mixed or unmixed streams), input T7 |
| IAFTBN=2 | (4) (1) | For afterburning (mixed or unmixed streams), input WFA |
| IDCD=1 | (3) | FAn duct nozzle will be convergent-divergent |
| IMCD=1 | (3) | Main nozzle will be convergent-divergent |
| NOZFLT=1 | (4) (3) | For floating main nozzle exit area |
| NOZFLT=2 | (4) (3) | For floating duct nozzle exit area |
| NOZFLT=3 | (4) (3) | For floating duct and main nozzle exit areas |
| ITRYS=N | | Number of passes through engine before quitting |
| TOLALL=X | | Tolerance which the errors must satisfy before engine is matched |
| DELFN, DELFG, DELSFC | | Normally input as 1.0 unless a correction is desired |
| ITITLE=1 | (1) | A title card must follow after the input data for this point (see below) |

A title card must be input immediately after the first point of the data pack and ITITLE must be set equal to 1 in the data for the first point. This is because a title is always printed for each point and must therefore be previously defined. The input format for the title is I2A6 and the resulting 72 spaces are centered on the page when printed out. The title may be changed by setting ITITLE=1 and inserting a new title card after the Namelist data for the point.

Off-Design Operation

Shown in the sample input listing are methods of specifying off-design operation points. The user inputs the appropriate control parameters, Mach number, altitude and power setting other than design values (power setting being a value for T4, PCNC, PCNF, or WFB).

If the engine has all its nozzles fixed, then an input such as T4 or shaft speed will set the thrust level. Other means of changing engine operation can be accomplished by varying nozzle throat areas A8 and A28. For example, an off-design condition may exist where the operating point lies outside the limits of the block data input for a component map such as the fan. A nozzle throat area change could return the operating point back on the input map. It should be noted that an area remains changed until it is recalculated by a new design case or altered by a new input.

The nozzle exit area (A9 and/or A29) may be floated to obtain full expansion of using NOZFLT=1, 2 or 3 for non-afterburning cases.

To run duct burning (fan stream only), cases load ETAD, and either T24 or WFD. To run afterburning, cases load ETAA, T7 or WFB. When such a point is run, the exhaust nozzle areas are allowed to float to obtain optimum expansion. This means that there can be no balancing at the point, and it is necessary to prebalance the engine cycle in a nonaugmented mode. That is, an identical point, except that it is nonaugmented must be run before either afterburning or duct burning. When either IAFTBN or IDBURN is greater than zero, the program will automatically set INIT=1 and use the balanced values from the preceding point. The nozzle areas are returned to their original values after completing an augmented point. Some examples of afterburning are given in the sample data listing.

SECTION VI
SUBROUTINE DESCRIPTIONS

A flow chart of the computer program with the subroutines is shown in Figure 5. Listed here are brief descriptions of the subroutines.

| | |
|--------|---|
| TRISPL | Dummy main program to initiate the calculations and cause the input of the controlled output variables. Because of the looping between subroutines, control is never transferred back to this routine. |
| ENGBAL | Main subroutine. Controls all engine balancing loops, checks tolerances and number of loops and loads matrix. |
| MATRIX | Solves error matrix |
| INPUT | Reads Namelist data and title. Prints title, |
| ZERO | Zeroes common and certain controls |
| COINLT | Determines ram recovery and performs inlet calculations |
| ATMOS | 1962 US Standard Atmosphere table |
| RAM | Calculates ram recovery defined by MIL-E-5008B Specifications |
| GUESS | Determines initial values of independent variables (PCNF, PCNI, PCNG, and T4) at each point. It may be desired to change these equations to suit a particular engine. The closer the initial values are to the final values, the faster the program will balance. |
| COFAN | Uses BLOCK DATA to perform fan calculations |
| COINTC | Uses BLOCK DATA to perform intermediate compressor calculations. |

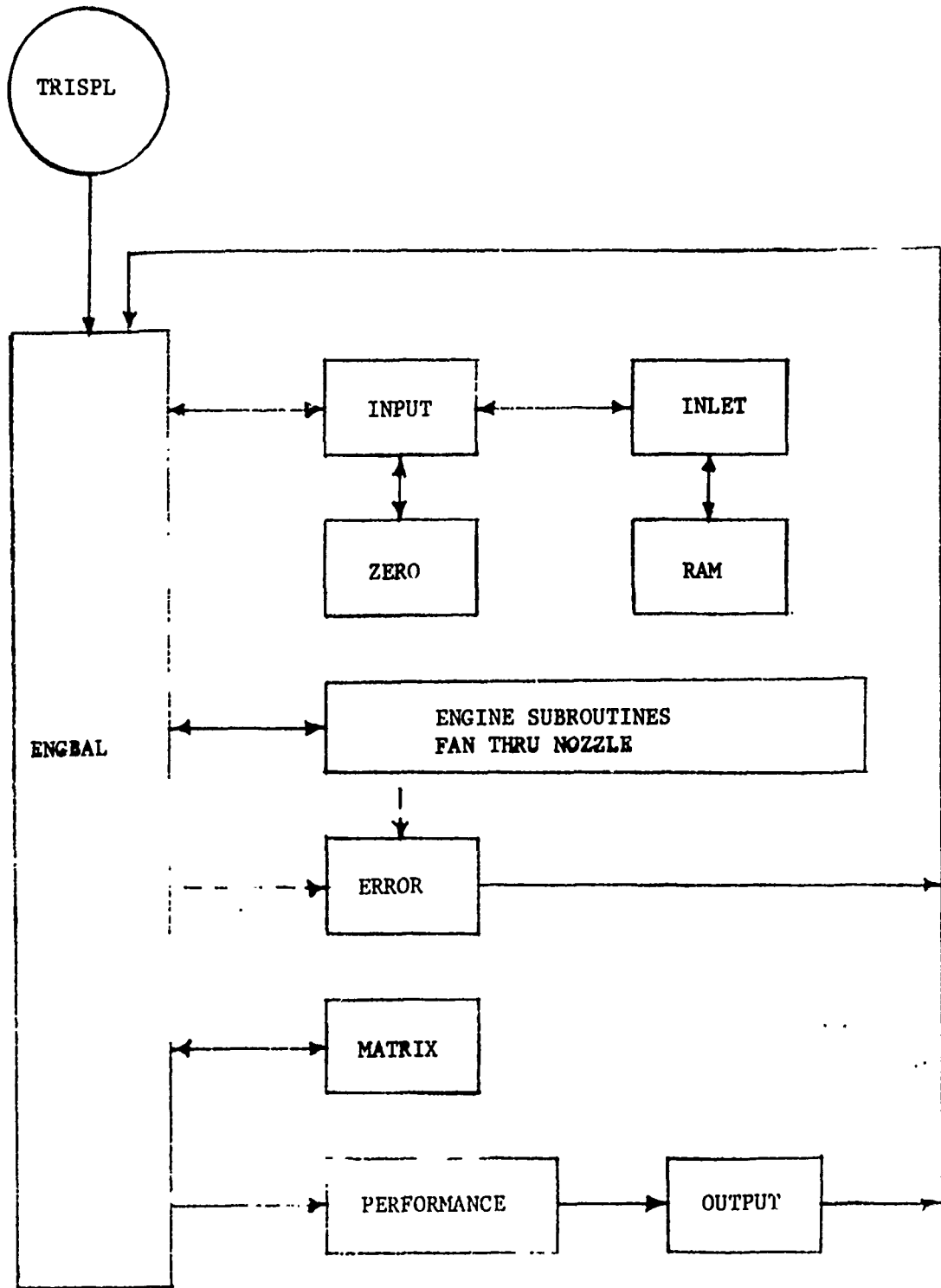


Figure 5. TRISPL Subroutine Flow Chart

COCOMP Uses BLOCK DATA to perform high-pressure compressor calculations. Calculates ERR(7).

COCOMB Uses BLOCK DATA to perform combustor calculations. May use either T4 or WFB as the main parameter.

COHPTB Uses BLOCK DATA to perform high-pressure turbine calculations. Calculates ERR(1) and ERR(2).

COIPTB Uses BLOCK DATA to perform intermediate turbine calculations. Calculates ERR(8) and ERR(9).

COLPTB Uses BLOCK DATA to perform low-pressure turbine calculations. Calculates ERR(3) and ERR(4).

FRTOSD Dummy routine to transfer values from common FRONT to common SIDE.

CODUCT Performs duct and duct-burning calculations. May use T24 or WFD as the main parameter for duct burning. Controls the duct nozzle and calculates ERR(7) if in separate-flow mode.

FASTBK Dummy routine to transfer values from common FRONT and SIDE to common BACK.

COMIX Performs gas mixing calculations if in mixed flow mode. At design points it calculates areas from either an input static pressure (PS55) or an input Mach number (AM55) if PS55=0. Also, an option exists where afterburner entrance area A6 is calculated as a function of an input afterburner entrance Mach number AM6 at the design point. At off-design points it calculates static

pressures and Mach numbers from the design areas.
Calculates ERR(5) if in mixed-flow mode.

COAFBN Performs afterburning calculations. May use either T7 or WFA as the main parameter.

COMNOZ Controls the main nozzle and calculates ERR(6).

PERF Calculates performance after engine is balanced.

OUTPUT Prints output except for controlled output. Prints the main commons in a close format after each point.

CONCUT Controls and prints the controlled output variables.

ERROR Controls all printouts if an error occurs. Prints name of subroutine where error occurred and also prints the values of all variables in the main commons.

SYG Controls printing from UNIT08. Throughout the program and particularly in ENGBAL, certain messages, variables, and matrix values are written on UNIT08 as an aid in determining why an error occurred or why a point did not balance. These values are printed out if subroutine ERROR is called and IDUMP is greater than zero, or after a good point if IDUMP=2.

TAPES Defines UNIT08, which is just a "scratch" disk and does not require a \$SETUP card. Normal input and output are on UNIT05 and UNIT06, respectively.

THCOMP Performs isentropic calculations for compressors.

THTURB Performs isentropic calculations for turbines.

THERMO Provides thermodynamic conditions using PROCOM.

| | |
|--------|---|
| PROCOM | Calculates thermodynamic gas properties for either air or a fuel-air mixture, based on JP-4. |
| SEARCH | General table lookup and interpolation routine to obtain data from Block Data routines. |
| MAPBAC | Used when calculations result in values not on the turbine maps. Changes the map value and an independent variable (PCNF, PCNC or T4) in an attempt to rectify the situation. |
| CONVRG | Performs nozzle calculations for a convergent nozzle. |
| CONDIV | Performs nozzle calculations for a convergent-divergent nozzle. |
| AFQUIR | General quadratic interpolation routine. |
| CMBDAT | Block Data for combustors. |
| FANDAT | Block Data for fan. |
| INTDAT | Block Data for intermediate compressor. |
| CMPDAT | Block Data for high-pressure compressor. |
| LTURB | Block Data for low-pressure turbine. |
| ITURB | Block Data for intermediate turbine. |
| HTURB | Block Data for high-pressure turbine. |

SECTION VII
PROGRAM LISTING

The following is a complete listing of all subroutines required
to run TRISPL.

```
PROGRAM TRISPL(INPUT,OUTPUT,TAPES=INPUT,TAPE6=OUTPUT,TAPE8)  
COMMON /POINT/ICATPT  
IDATPT=0  
CALL CONOUT(1)  
CALL ENGBAL  
STOP  
END
```

```

SUBROUTINE RGBAL
COMMON / AGL/
1WORD ,IDES ,JOES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX ,IDURN ,IAFTBN ,IDCD ,INCD ,IOSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(9)
COMMON/DESIGN/
1PCNFQU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF
3ZI DS ,PCNI DS ,PRI DS ,ETAIDS ,WAIDS ,FRICF ,ETAICF ,WAICF
4ZC DS ,PCNC DS ,PRC DS ,ETA CDS ,WAC DS ,PRCCF ,ETA CCF ,WACCF
5T4 DS ,WF8 DS ,DTC DS ,ETAB DS ,WA3C DS ,DPC DS ,DTC CCF ,ETABCF
6TFHP DS ,CNHP DS ,ETHP DS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T20S
7TFIP DS ,CNIF DS ,EY1P DS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T210S
8TFLP DS ,CNLP DS ,ETLP DS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T220S
9T24 DS ,WF0 DS ,OTDU DS ,ETA DS ,WA23 DS ,OPDU DS ,OTDUCF ,ETA DCF
AT7 DS ,WFA DS ,OTA FDS ,ETA DS ,WG6C DS ,OPA FDS ,OTA FCF ,ETA ACF
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU
6CNF ,PRF ,ETA F ,WAF C ,WAF ,BLDUI ,BLDUC ,BLOB
7CNI ,PRI ,ETA I ,WAIC ,WAI ,BLOBI ,BLOB C ,WA3
8CNC ,PRC ,ETA C ,WACC ,WAC ,ETAB ,OPCOM ,WG4
9CNHP ,ETA THP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4
ACNIP ,ETA TIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF
BCNLP ,ETA TLF ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS
CNG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM
DALTP ,ETA R ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC
EWFB ,TFFH ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLOUI
FPCBLDUC ,PCBLOBI ,PCBLOB C ,PCBLHPI ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLLPI
GPCBLLFC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XBLF ,XBLOU ,XBLOUI ,XBLOUC
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6
6WAD ,WFD ,WG24 ,FAR24 ,ETAU ,DPOUC ,BYPASS ,DUMS7
7TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XFP1 ,DUMB
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9
5WG6 ,WFA ,WG7 ,FAR7 ,ETA A ,OPA FT ,V55 ,V25
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9
8VA ,FRD ,VJD ,FGHD ,VJM ,FGHM ,FGPD ,FGPM
9VGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DIMENSION DELSAV(9)
DIMENSION VAR(9),DEL(9),ERRB(9),DELVAR(9),EMAT(9,9),VMAT(9),

```

```

1 AMAT(9)
  DATA DELSA V/9*.001/
  DATA AWORD/6HENG BAL/
  DATA VDELTA, VLI P, VCHNGE, NOMISX/
1 0.001, 0.100, 0.850, 4/
  DATA DEL/9*0./
  CALL INPUT
  IF (INIT.EQ.1) GO TO 50
  TFFHP=TFHPDS
  TFFIP=TFIPDS
  TFFLP=TFLPDS
50  LOOPER=0
  NUMMAP=0
  NOMISS=0
1  LOOP=0
  MISMAT=0
  NOMAP=0
  IGO=2
  DO 2 I=1,9
    VMAT(I)=0.
    AMAT(I)=0.
    DELVAR(I)=0.
    DO 2 L=1,9
2  EMAT(I,L)=0.
3  LOOPER=LOOPER+1
  CALL COFAN
  WORD=AWORD
  IF (LOOPER.GT.ITRYS) GO TO 20
  IF (NOMAP.GT.0) GO TO 1
  NUMMAP=0
55  VAR(1)=ZF*100.
  IF (MODE.NE.3) VAR(2)=PCNF
  IF (MODE.EQ.3) VAR(2)=T4/10.
  VAR(3)=ZC*100.
  IF (MODE.NE.1) VAR(4)=PCNC
  IF (MODE.EQ.1) VAR(4)=T4/10.
  VAR(5)=TFFHP
  VAR(6)=TFFLP
  VAR(7)=ZI*100.
  VAR(8)=PCNI
  VAR(9)=TFFIP
  DO 4 I=1,9
  IF (ABS(ERR(I)).GT.TOTALL) GO TO 5
4  CONTINUE
  CALL PERF
  CALL ERROR
5  IF (LOOP.GT.0) GO TO 7
  MAPEOG=0
  MAPSET=0
  DO 6 I=1,9
  ERRO(I)=ERR(I)
6  DEL(I)=VDELTA*VAR(I)
  GO TO 9
7  IF (MISMAT.GT.0) GO TO 30
  IF (MAPEOG.EQ.0) GO TO 70

```

```

MAPEDG=0
MAPSET=1
VAR(LOOP)=VAR(LOOP)+2.*DEL(LOOP)
GO TO 10
70 IF (MAPSET.EQ.0) VAR(LOOP)=VAR(LOOP)+DEL(LOOP)
IF (MAPSET.EQ.1) VAR(LOOP)=VAR(LOOP)-DEL(LOOP)
MAPSET=0
DO 8 I=1,9
IF (DEL(LOOP).NE.0.) DELSAV(LOOP)=DEL(LOOP)
IF (DEL(LOOP).EQ.0.) DEL(LOOP)=DELSAV(LOOP)
8 EMAT(I,LOOP)=(ERRB(I)-ERR(I))/DEL(LOOP)
9 LOOP=LOOP+1
IF (LOOP.GT.9) GO TO 11
VAR(LOOP)=VAR(LOOP)-DEL(LOOP)
10 ZF=VAR(1)/100.
IF (MODE.NE.3) PCNF=VAR(2)
IF (MODE.EQ.3) T4=VAR(2)*10.
ZC=VAR(3)/100.
IF (MODE.NE.1) PCNC=VAR(4)
IF (MODE.EQ.1) T4=VAR(4)*10.
TFFHP=VAR(5)
TFFLP=VAR(6)
ZI=VAR(7)/100.
PCNI=VAR(8)
TFFIP=VAR(9)
IF (ZF.LT.0.) ZF=0.05
IF (ZI.LT.0.) ZI=0.05
IF (ZC.LT.0.) ZC=0.05
GO TO (1,3),IGO
11 DO 12 I=1,9
12 AMAT(I)=-ERRB(I)
DO 14 I=1,9
IZERO=0
DO 13 LOOP=1,9
13 IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1
IF (IZERO.LT.9) GO TO 14
WRITE(6,100) I
LOOPER=ITRYS+100
GO TO 20
14 CONTINUE
DO 16 LOOP=1,9
IZERO=0
DO 15 I=1,9
15 IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1
IF (IZERO.LT.9) GO TO 16
WRITE(6,101) LOOP
LOOPER=ITRYS+100
GO TO 20
16 CONTINUE
17 CALL MATRIX(EMAT,VMAT,AMAT)
LBIG=0
VARBIG=0.0
DO 18 L=1,9
ABSVAR=ABS(VMAT(L))
IF (ABSVAR.LE.VLIN*VAR(L)) GO TO 18

```

```

IF (ABSVAR.LE.VARBIG) GO TO 18
LBIG=L
VARBIG=ABSVAR
18 CONTINUE
VRATIO=1.0
IF (LBIG.GT.0) VRATIO=VLIM*VAR(LBIG)/VARBIG
ERRAVE=0.0
VMTAVE=0.0
DELAVE=0.0
DO 19 L=1,9
DELVAR(L)=VRATIO*VMAT(L)
ERRAVE=ERRAVE+ABS(AMAT(L))
VMTAVE=VMTAVE+ABS(VMAT(L))
DELAVE=DELAVE+ABS(DELVAR(L))
19 VAR(L)=VAR(L)+DELVAR(L)
ERRAVE=ERRAVE/9.
VMTAVE=VMTAVE/9.
DELAVE=DELAVE/9.
IF (MISMAT.GT.0) GO TO 32
IF (NOMISS.EQ.0) MISMAT=1
IF (MISMAT.EQ.0) IGO=1
20 WRITE(8,102) LOOPER
DO 21 I=1,9
21 WRITE(8,103) AMAT(I),(EMAT(I,L),L=1,9),VMAT(I),DELVAR(I),VAR(I)
WRITE(8,104) ERRAVE,VMTAVE,DELAVE
22 IF (LOOPER.LT.ITRYS) GO TO 10
CALL ERROR
RETURN
30 VMTAVX=VMTAVE
DO 31 I=1,9
31 AMAT(I)=-ERR(I)
GO TO 17
32 WRITE(8,105) AMAT,ERRAVE,DELVAR,DELAVE,VMAT,VMTAVE,VAR
MISMAT=MISMAT+1
IF (VMTAVE.LT.VCHNGE*VMTAVX) GO TO 22
WRITE(8,105)
IF (MISMAT.LT.NOMISS) NOMISS=1
MISMAT=0
LOOP=0
IGO=2
GO TO 55
100 FORMAT(4H3ROW,I2,16H IS ZERO IN EMAT)
101 FORMAT(7HCOLUMN,I2,16H IS ZERO IN EMAT)
102 FORMAT(7H0 ERRB,31X23HERROR MATRIX AFTER LOOP,I4,31X4HVMAT,
16X6HDELVAR,7X10HVARIABLE$$)
103 FORMAT(1H0,F7.3,9F9.3,4XF9.3,3XF10.4,4XF11.4)
104 FORMAT(1H0,F8.4,32X14HAVERAGE VALUES,31X,2F11.4,5H$$$$$$)
105 FORMAT(12H0----- AMAT,10F11.6,6H$$$$$$,
1/, 12H -----DELVAR,10F11.6,6H$$$$$$,
2/, 12H ----- VMAT,10F11.6,6H$$$$$$,
3/, 12H ----- VAR,9F11.6,6H$$$$$$)
106 FORMAT(1H0,56X22HCHANGE TOO SMALL$$$$$$)
END

```

```

SUBROUTINE INPUT
DIMENSION TITLE(12)
COMMON /POINT/IDATPT
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAHTP ,
2IGASMX,IOBURN,IAFTBN,IDCD ,IMCD ,IOSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOMAP,NUMMAP,MAPEDG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIOS ,PCNIOS ,PRIDS ,ETAIDS ,WAIOS ,PRICF ,ETAICF ,WAICF ,
4ZCOS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3COS ,DPCODS ,DTCOCF ,ETABCF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHMPCF ,T20S ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T210S ,
8TFLPDS ,CNLPDS ,ETL PDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T220S ,
9T24DS ,WFODS ,DTCODS ,ETAADS ,WA23CS ,DPOUDS ,DTCOCF ,ETAOCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
BA5S ,A2S ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/FRCNT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETA F ,WAF C ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETA I ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETA C ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9CNHP ,ETA THF ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETA TIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETA TLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CNG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETA R ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FFCBLDUC ,PCBL CBI ,PCBL OBC ,PCBLHPI ,PCBLHPC ,PCBLIFI ,PCBLIPC ,PCBL LPI ,
GPCBL LPC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XBLF ,XBLOU ,XBLOUI ,XBLOUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6HAD ,WFC ,WG24 ,FAR24 ,ETA D ,DPOUC ,BYPASS ,DUMS7 ,
7TS28 ,FS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXHFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XFP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,HFA ,WG7 ,FAR7 ,ETA A ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPO ,FGFM ,
9FGH ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC

```



```

NAMELIST/DATAIN/ IEND,
AITITLE , IDES , MODE , INIT , ICUMP , IAMTP , IGASM ,
BIOBURN , IAFTBN , IDCO , IMCO , NCZFLT , ITRYS , TOLALL ,
CZFDOS , PCNFOS , PRFOS , ETAFOS , WAFOS , PRFCF , ETAFCF , WAFCF ,
DZIOS , PCNIOS , PRIOS , ETAIOS , WAIDS , PRICF , ETAICF , WAICF ,
EZCDS , PCNCOS , FRCOS , ETACOS , , PRCCF , ETACCF , WACCF ,
FT4OS , WFEDS , OTCOS , ETABOS , WA3COS , OPCOS , OTCOCF , ETABCF ,
GTFHPDS , CNHPDS , ETHPDS , TFHPCF , CNHPCF , ETHPCF , OHHPCF , T2OS ,
HTFIPDS , CNIPDS , ETIPDS , TFIPCF , CNIPCF , ETIPCF , OHIPCF , T21OS ,
ITFLPCS , CNLPCS , ETLPOS , TFLPCF , CNLPCF , ETLPCF , DHLPCF , T22OS ,
JT24OS , WFUDS , OTDUDS , ETAODS , WA23OS , OPDUDS , OTDUCF , ETAOCF ,
KT7OS , WFAOS , OTAFOS , ETAADS , WG6COS , OPAFOS , OTAFCF , ETAACF ,
LA55 , A25 , A6 , A7 , A8 , A9 , A28 , A29 ,
MT2 , P2 , T4 , ZF , PCNF , ZI , ZC , PCNC ,
NWF8 , TFFHP , TFFIP , TFFLP , AM , ALTP , ETAR , HPEXT ,
OPCBLF , BLF , PCBLI , BLI , PCBLC , BLC , PCBLOUI , PCBLOUC ,
PBLDUI , BLDUC , PCBLOBI , PCBLOBC , BLOBI , BLOBC , PCBLHPI , PCBLHPC ,
QPCBLIPI , PCBLIPC , PCBLLFI , PCBLLPC , BLHPI , BLHPC , BLIFI , BLIPC ,
RELLPI , BLLPC , PS55 , AM55 , AM6 , T24 , ETAD , WFO ,
ST7 , ETAA , WFA , CVNOZ , CVMNOZ , DELFG , DELFN , DELSFC

```

```

DATA IEND/0/
C *** ITITLE=1 WILL READ IN TITLE
C *** IDES =1 FOR CALCULATING DESIGN POINT
C *** MODE =0 FOR CONSTANT T4
C *** MODE =1 FOR CONSTANT PCNC
C *** MODE =2 FOR CONSTANT WFB
C *** MODE =3 FOR CONSTANT PCNF
C *** INIT =1 WILL NOT INITIALIZE POINT
C *** IDUMP =1 WILL DUMP LOOPING WRITE-OUTS IF ERRCR OCCURS
C *** IDUMP =2 WILL DUMP LOOPING WRITE-OUTS AFTER EVERY POINT
C *** IAMTP =0 WILL USE INPUT AM AND MIL SPEC ETAR
C *** IAMTP =1 WILL USE INPUT AM AND INPUT ETAR
C *** IAMTP =2 WILL USE T2 AS T1=T1+T2 AND STANDARD P1
C *** IAMTP =3 WILL USE P2 AND STANDARD T1
C *** IAMTP =4 WILL USE T2 AND P2
C *** IGASM=0 SEPARATE FLOW, A6=A55
C *** IGASM=1 WILL MIX DUCT AND MAIN STREAMS, A6=A25+A55
C *** IGASM=2 WILL MIX DUCT AND MAIN STREAMS, INPUT AM6
C *** IDURN=1 FOR DUCT BURNING, INPUT T24
C *** IDURN=2 FOR DUCT BURNING, INPUT WFO
C *** IAFTBN=1 FOR AFTERBURNING, INPUT T7
C *** IAFTBN=2 FOR AFTERBURNING, INPUT WFA
C *** IDCO =1 DUCT NOZZLE WILL BE C-0
C *** IMCO =1 MAIN NOZZLE WILL BE C-0
C *** NOZFLT=1 FOR FLOATING MAIN NOZZLE
C *** NOZFLT=2 FOR FLOATING DUCT NOZZLE
C *** NOZFLT=3 FOR FLOATING MAIN AND DUCT NOZZLES
C *** ITRYS =N NUMBER OF PASSES THRU ENGINE BEFORE QUITTING
DATA AWORD/6H INPUT/
IDATPT=ICATPT+1
CALL ZERO
WORD=AWORD
READ(5,DATAIN)
IF(IEND.NE.0) STOP
CALL REMARK(14HNEW DATA POINT)

```

```

      IF (IAFTBN.GT.0.OR.IDBURN.GT.0) INIT=1
      IF (ITITLE.EQ.1) READ(5,101) TITLE
      ITITLE=0
      WRITE(6,102) TITLE
      IF (MODE.EQ.0) WRITE(8,103) IDES,AM,ALTP,T4 ,T24,T7
      IF (MODE.EQ.1) WRITE(8,104) IDES,AM,ALTP,PCNC,T24,T7
      IF (MODE.EQ.2) WRITE(8,105) IDES,AM,ALTP,WFB ,T24,T7
      IF (MODE.EQ.3) WRITE(8,106) IDES,AM,ALTP,PCNF,T24,T7
      CALL COINLT
      RETURN
101  FORMAT(12A6)
102  FORMAT(1H1,30X12A6)
103  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F8.2,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
104  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H PCNC=,F8.3,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
105  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H WFB=,F8.4,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
106  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H PCNF=,F8.3,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
      END

```

```

SUBROUTINE ZERO
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX, IDBURN, IAFTBN, IDCD ,IMCD ,IDSHOC, IMSHOC, NOZFLT,
3ITRYS, LOOPER, NOMAP, NUMMAP, MAPEDG, TOLALL, ERR(9)
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CHG45 ,FAR45 ,HG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLDUC, PCBLCBI, PCBLOBC, PCBLHPI, PCBLHPC, PCBLIPI, PCBLIPC, PCBLLPI,
GPCBLLPC

```

```

COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XBLF ,XBLDU ,XBLDUI ,XBLDUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6WAD ,WFD ,WG24 ,FAR24 ,ETAO ,DPOUC ,BYPASS ,DUMS7 ,
7TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29

```

```

COMMON / BACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXWFB ,XWG55 ,XFAR55, XWFD ,XWG24 ,XFAR24, XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC

```

```

DIMENSION Z1(94),Z2(56),Z3(72)

```

```

EQUIVALENCE (Z1(1),T1), (Z2(1),XP1), (Z3(1),XT55)

```

```

DATA ISTART/0/

```

```

ISTART=ISTART+1

```

```

IDES=0

```

```

JDES=0

```

```

INIT=0

```

```

IDBURN=0

```

```

IAFTBN=0

```

```

IDSHOC=3

```

```

IMSHOC=3

```

```

IF(ISTART.NE.1) GO TO 4

```

```

C*** ZERO FRONT, SIDE, AND BACK INITIALLY TO PREVENT CDC 6600 MODE ERRORS

```

```

DO 5 I=1,94

```

```

5 Z1(I)=0.

```

```
DO 6 I=1,56
6 Z2(I)=0.
DO 7 I=1,72
7 Z3(I)=0.
CALL SYG(1)
RETURN
4 CONTINUE
T2Q=T2
P2Q=P2
T4Q=T4
DO 1 I=1,94
1 Z1(I)=0.
DO 2 I=1,56
2 Z2(I)=0.
DO 3 I=1,72
3 Z3(I)=0.
T2=T2Q
P2=P2Q
T4=T4Q
CALL SYG(1)
RETURN
END
```

SUBROUTINE CCINLT

COMMON / ALL/

1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
 2 IGASHX, IDBURN, IAFTBN, IOCD ,IMCD ,IDSHOC, IMSHOC, NOZFLT,
 3 IT RYS, LOOPER, NOHAP, NUMHAP, HAPEDG, TOLALL, ERR(9)

COMMON/DESIGN/

1 PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
 2 ZFDS ,PCNFDS ,PRFDS ,ETAZDS ,MAZDS ,PRFCF ,ETAFCF ,MAFCF ,
 3 ZIDS ,PCNIDS ,PRIIDS ,ETAIDS ,MAIDS ,PRICF ,ETAICF ,MAICF ,
 4 ZCDS ,PCNCDS ,PRCDS ,ETAZDS ,MACDS ,PRCCF ,ETAACF ,MACCF ,
 5 T4DS ,WFBS ,OTCDS ,ETABDS ,MA3CDS ,DPCDS ,DTCOCF ,ETABCF ,
 6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
 7 TFIPDS ,CNIFDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
 8 TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
 9 T24DS ,WFDS ,OTDUDS ,ETAADS ,MA23DS ,OPDUDS ,OTDUCF ,ETAACF ,
 AT7DS ,WFADS ,DTAFDS ,ETAADS ,MG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
 8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
 CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV

COMMON/ FRCNT/

1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
 2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
 3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
 4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
 5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
 6 CNF ,PRF ,ETAF ,MAFC ,MAF ,BLOUI ,BLOUC ,BLOB ,
 7 CNF ,PRI ,ETAI ,MAIC ,MAI ,BLOBI ,BLOBC ,WA3 ,
 8 CNC ,PRC ,ETAC ,MACC ,MAC ,ETAB ,DPCOM ,MG4 ,
 9 CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
 ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
 BCNLP ,ETATLP ,DHTCLP ,DHTC ,BLLP ,BLLPI ,BLLPC ,CS ,
 C*G45 ,FAR45 ,MG5 ,FAR5 ,MG55 ,FAR55 ,HPEXT ,AM ,
 DALTP ,ETAR ,ZF ,PCN ,ZI ,PCNI ,ZC ,PCNC ,
 EWF8 ,TFHFP ,TFHFP ,TFHFP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
 FFCBLOUC, PCBLOBI, PCBLOBC, PCBLHPI, PCBLHPC, PCBLIPI, PCBLLPC, PCBLLPI,
 GPCBLLPC

DATA AWORD/6HCOINLT/

WORD=AWORD

AJ=778.26

G=32.174049

ALT=ALTP*2.0855531E+07/(2.0855531E+07-ALTP)

CALL ATMOS(ALT, T1, XX1, XX2, XX3, P1, CS, XX4, IIER)

IF (IAMTP.EG.2) T1=T1+T2

1 IF (IAMTP.NE.1) CALL RAM(AM, ETAR)

FAR=0.0

CALL FROCOM(FAR, T1, CS, XX2, XX3, R1, PHI1, H1)

S1=PHI1-R1*ALOG(P1)

H2=H1+(AM*CS)**2/(2.*AJ*G)

P2T=1.

DO 2 I=1,10

CALL THERMC(P2T, H2, T2T, S2T, AM, 0, 0.0, 1)

IF (ABS(S2T-S1).LE.0.0001*S1) GO TO 3

2 P2T=P1*EXP((AM/1.986375)*((S2T-S1)+(1.986375/AM)*ALOG(P2T/P1)))

CALL ERROR

RETURN

3 IF (IAMTP.EG.3.OR.IAMTP.EQ.4) ETAR=P2/P2T

```
P2=ETAR*P2T
IF (IAMTP.NE.4) CALL THERMO(P2,H2,T2,S2,XX5,C,0.0,1)
IF (IAMTP.EQ.4) CALL THERMO(P2,H2,T2,S2,XX5,0,0.0,0)
IF (INIT.EQ.1) GO TO 6
IF (IDES.EQ.1) GO TO 4
IF (MODE.EQ.3) GO TO 5
PCNF=GUESS (HCOE,T4,T4DS,PCNC,PCNCDS,WFB,WFBDS,T2,T2DS,PCNFOS)
PCNFGU=PCNF
GO TO 5
4 PCNF=PCNFOS
  PCNFGU=PCNF
  T2DS=T2
5 ZF=ZFOS
6 RETURN
END
```

```

SUBROUTINE ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)
THIS IS A SUBROUTINE TO COMPUTE CERTAIN ELEMENTS OF THE 1962
U.S. STANDARD ATMOSPHERE UP TO 90 KILOMETERS.
CALLING SEQUENCE...

CALL ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)
ZFT = GEOMETRIC ALTITUDE (FEET)
TM = MOLECULAR SCALE TEMPERATURE (DEGREES RANKINE)
SIGMA = RATIO OF DENSITY TO THAT AT SEA LEVEL
RHO = DENSITY (LB-SEC**2-FT**(-4) OR SLUGS-FT**3)
THETA = RATIO OF TEMPERATURE TO THAT AT SEA LEVEL
DELTA = RATIO OF PRESSURE TO THAT AT SEA LEVEL
CA = SPEED OF SOUND (FT/SEC)
AMU = VISCOSITY COEFFICIENT (LB-SEC/FT**2)

K = 1 NORMAL
   = 2 ALTITUDE LESS THAN -5000 METERS OR GREATER THAN 90 KM
   = 3 FLOATING POINT OVERFLOW

ALL DATA AND FUNDAMENTAL CONSTANTS ARE IN THE METRIC SYSTEM AS
THESE QUANTITIES ARE DEFINED AS EXACT IN THIS SYSTEM.

THE RADIUS OF THE EARTH (REFT59) IS THE VALUE ASSOCIATED WITH THE
1959 ARDC ATMOSPHERE SO THAT PROGRAMS CURRENTLY USING THE LIBRARY
ROUTINE WILL NOT REQUIRE ALTERATION TO USE THIS ROUTINE.
DIMENSION HB(10),TMB(10),DELTAB(10),ALM(10)
DATA (HB(I),I=1,10)/-5.,0.,11.,20.,32.,47.,52.,61.,79.,88.743/
DATA (TMB(I),I=1,10)/320.65,288.15,216.65,216.65,228.65,270.65,
1 270.65,252.55,180.65,180.65/
DATA (DELTAB(I),I=1,10)/1.75363,1.,2.23361E-01,5.40328E-02,
1 8.56663E-03,1.49455E-03,5.82289E-04,1.79718E-04,1.0241E-05,
2 1.6223E-06/
DATA (ALM(I),I=1,10)/-6.5,-6.5,0.,1.,2.8,2.,-2.,-4.,0.,0./
DATA REFT59/2.0855531E 07/, GZ /9.80665/,
A AMZ /28.9644 /, RSTAR /8.31432/,
B FTOKM/3.048E-04 /, S /110.4 /,
C AMUZ /1.2024E-05 /, CAZ /1116.45/,
D RHOZ /0.076474 /, GZENG /32.1741/

CONVERT GEOMETRIC ALTITUDE TO GEOPOTENTIAL ALTITUDE
HFT = (REFT59/(REFT59+ZFT))*ZFT
CONVERT HFT AND ZFT TO KILOMETERS
Z = FTOKM*ZFT
H = FTOKM*HFT
K = 1
TMZ = TMB(Z)
IF (H.LT.-5.0.OR.Z.GT.90.0) GO TO 16
DO 10 M=1,10
IF (H-HB(M)) 11,12,10
10 CONTINUE
GO TO 16
11 M = M-1
12 DELH = H-HB(M)
IF (ALM(M).EQ.0.0) GO TO 13
TMK = TMB(M)+ALM(M)*DELH
GRADIENT IS NON ZERO, PAGE 10, EQUATION I.2.10-(3)

```

```

      DELTA = DELTAB(M)*((TMB(M)/TMK)**(GZ*AMZ/(RSTAR*ALM(M))))
      GO TO 14
13  TMK = TMB(M)
C   GRADIENT IS ZERO, PAGE 10, EQUATION I.2.10-(4)
      DELTA = DELTAB(M)*EXP(-GZ*AMZ*DELH/(RSTAR*TMB(M)))
14  THETA = TMK/THZ
      SIGMA = DELTA/THETA
      ALPHA = SQRT(THETA**3)*((TMZ+S)/(TMK+S))
C   CONVERSION TO ENGLISH UNITS
      TM = 1.8*TMK
      RHO = RHOZ*SIGMA/GZENG
      CA = CAZ*SQRT(THETA)
      AMU = AMUZ*ALPHA/GZENG
C   CALL OVERFL(J)
      J=2
      GO TO (15,17), J
15  K = K+2
      GO TO 17
16  K = 2
17  RETURN
      END

```


SUBROUTINE COFAN

COMMON / ALL/

1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAHPT ,
 2 IGASHX, IUBURN, IAFTBN, IDCD ,IMCD ,IDSHOC, IMSHOC, NOZFLT,
 3 ITRYS, LOOPER, NOMAP, NUMMAP, MAPELG, TOLALL, ERR(9)

COMMON/DESIGN/

1 PCNFGU ,PCNIGU ,FCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
 2 ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
 3 ZIDS ,PCNICS ,FRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
 4 ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
 5 T4DS ,WF8DS ,OTCDS ,ETABDS ,WA3CDS ,OPCDS ,OTCCF ,ETABCF ,
 6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
 7 TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
 8 TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
 9 T24DS ,WFCDS ,OTDUDS ,ETADDS ,WA23DS ,OPDUDS ,DTOUFCF ,ETADCF ,
 A T7DS ,WFAOS ,DTAFDS ,ETAADS ,MG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
 BA55 ,A25 ,A6 ,A7 ,A6 ,A9 ,A28 ,A29 ,
 CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV

COMMON/FRONT/

1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
 2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
 3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
 4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
 5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
 6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
 7 CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
 8 CNC ,FRC ,ETAC ,WACC ,WAC ,ETAB ,OPCCM ,MG4 ,
 9 CNHP ,ETATHP ,DHTGHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
 ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
 BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLLC ,CS ,
 CWG45 ,FAR45 ,MG5 ,FAR5 ,MG55 ,FAR55 ,HPEXT ,AM ,
 DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
 EMFR ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
 PCBLOUC, PCBLCBI, PCBLCBC, PCBLCPI, PCBLCPC, PCBLCPI, PCBLCPC, PCBLLPI,
 PCBLLPC

COMMON / FAN/CNX(15), PRX(15,15), WACX(15,15), ETAX(15,15),

1 NCN, NPT(15)

DIMENSION WLN(2)

DATA ANORD, WLN/6H COFAN, 6H (LO) ,6H (HI) /

WORD=ANORD

THETA=SQRT(12/518.658)

IF (IDES.NE.1) GO TO 10

THETA0=THETA

10 CNF=PCNF*THETA0/(100.*THETA)

IF (ZF.LE.0.) ZF=0.

IF (ZF.GT.1.) ZF=1.

CNFS=CNF

CALL SEARCH(ZF,CNF,PRF,WAF,ETAF,

1 CNX(1),NCN,PRX(1,1),WACX(1,1),ETAX(1,1),NPT(1),15,15,IGO)

IF ((CNF-CNFS).GT.0.0005*CNF) MAPELG=1

IF (IGO.EQ.1.OR.IGO.EQ.2) WRITE(8,1000) CNFS,WLN(IGO)

1000 FORMAT(19H0* * * CNF OFF MAP,F10.4,2X A6,11H* * *\$\$\$\$\$)

WAF=WAF*P2/THETA

IF (IDES.NE.1) GO TO 1

PRFCF=(PRFCS+1.)/(PRF-1.)

```

      ETAFCF=LTAFDS/ETAF
      WAFCF=WAFDS/WAF
      WRITE(6,100) PRFCF,ETAFCF,WAFCF,T2DS
100  FORMAT(11HOFAN DESIGN,13X8H PRFCF=,E15.8,8H ETAFCF=,E15.8,
18H WAFCF=,E15.8,8H T2DS=,E15.8)
1    PRF=PRFCF*(PRF-1.)+1.
      ETAF=ETAFCF*ETAF
      WAF=WAFCF*WAF
      CALL THCOMP(PRF,ETAF,T2,H2,S2,P2,T21,H21,S21,F21)
      IF(PCBLF.GT.0.) BLF=PCBLF*WAF
      IF(JDES.EQ.1) GO TO 7
      JDES=1
      IF(INIT.EQ.1) GO TO 6
      IF(IDES.EQ.1) GO TO 4
      IF(MODE.NE.2) GO TO 2
      T4=GUESS(3,Y1,Y2,PCNF,PCNFDS,WFB,WBDS,Y7,Y8,T4DS)
      PCNI=GUESS(8,T4,T4DS,Y3,Y4,Y5,Y6,T21,T21DS,PCNIDS)
      PCNC=GUESS(4,Y1,Y2,PCNF,PCNFDS,WFB,WBDS,Y7,Y8,PCNCDS)
      GO TO 5
2    IF(MODE.EQ.1) GO TO 3
      IF(MODE.EQ.9) GO TO 20
      T4=GUESS(7,Y1,Y2,PCNF,PCNFDS,Y5,Y6,T2,T2DS,T4DS)
20   PCNC=GUESS(5,T4,T4DS,Y3,Y4,Y5,Y6,T21,T21DS,PCNCDS)
      PCNI=GUESS(9,Y1,Y2,PCNC,PCNCDS,Y5,Y6,T21,T21DS,PCNIDS)
      GO TO 5
3    T4=GUESS(6,Y1,Y2,PCNC,PCNCDS,Y5,Y6,T21,T21DS,T4DS)
      PCNI=GUESS(8,T4,T4DS,Y3,Y4,Y5,Y6,T21,T21DS,PCNIDS)
      GO TO 5
4    PCNC=PCNCDS
      PCNI=PCNIDS
      T4=T4DS
      WFB=WBDS
      T21DS=T21
5    ZC=ZCDS
      ZI=ZIDS
      PCNIGU=PCNI
      PCNCGU=PCNC
      T4GU=T4
6    INIT=0
7    IF(MODE.NE.3) GO TO 8
      IF(ABS(CNF-CNFS).LE.0.001*CNFS) GO TO 9
      WRITE(8,200) CNFS,CNF
200  FORMAT(10HCNF WAS= ,E15.8,11H AND NOW= ,E15.8,
124H CHECK PCNF INPUTS$$$$)
      CALL ERROR
8    PCNF=100.*THETA*CNF/THETA0
9    CALL COINTC
      RETURN
      END

```

```

SUBROUTINE COINTC
COMMON/ ALL/
1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IANTP ,
2 IGASMX ,IDURN ,IAFTBN ,IDCD ,IMCO ,IDSHGC ,IMSHGC ,NOZFLT ,
3 IT RYS ,LOOPER ,NONAP ,NUMMAP ,NAPEGG ,TOLALL ,ERR (9)
COMMON/DESIGN/
1 PCNFGU ,PCNIGU ,PCNCCU ,T4GU ,DLMD1 ,DELFG ,DEIFN ,DELSFC
2 ZFDS ,PCMFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF
3 ZIDS ,PCNICS ,FRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF
4 ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF
5 T4DS ,WFEDS ,DTGDS ,ETABDS ,WACDS ,DPCDS ,DTGDCF ,ETABCF
6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHPHPCF ,T2DS
7 TFIPDS ,CNIPDS ,EYIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHPHPCF ,T2IDS
8 TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHPHPCF ,T2IDS
9 T24DS ,WFADS ,DTAADS ,ETAADS ,WACDS ,DPAADS ,DTAADF ,ETAADF
A7DS ,WFADS ,DTAADS ,ETAADS ,WACDS ,DPAADS ,DTAADF ,ETAADF
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29
CPS55 ,AM55 ,CVNOZ ,CVNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/FRONT/
1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2
2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22
3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4
4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5
5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU
6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB
7 CNJ ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3
8 CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCCH ,WG4
9 CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC
EWFB ,TFHFP ,TFHFP ,TFHFP ,PCBLF ,PCBLI ,PCBLC ,PCBLG
FPCBLDUC ,PCBLCBI ,PCBLCBC ,PCBLHFZ ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLLE
GPCBLLE
COMMON/IPHP/WA22
COMMON/ INT/CNX (15) ,PRX (15,15) ,WACX (15,15) ,ETAX (15,15) ,
1 NCN ,NPT (15)
DIMENSION WLH (2)
DATA AWORD ,WLH/6HCOINTC ,6H (LO) ,6H (HI) /
WORD=AWORD
THETA=SQRT (T21/518.668)
IF (IDES.NE.1) GO TO 10
THETAQ=THETA
10 CNI=PCNI*THETAQ/(100.*THETA)
IF (ZI.LT.0.) ZI=0.
IF (ZI.GT.1.) ZI=1.
CNIS=CNI
CALL SEARCH (ZI ,CNI ,PRI ,WAIC ,ETAX ,
1 CNX (1) ,NCN ,PRX (1,1) ,WACX (1,1) ,ETAX (1,1) ,NPT (1) ,15,15 ,IGO)
IF ((CNI-CNIS).GT..0005*CNI) NAPEGG=1
IF (IGO.EQ.1.OR.IGO.EQ.2) WRITE (8,1000) CNIS ,WLH (IGO)
1000 FORMAT (19H0* * * CNI OFF MAP ,F10.4 ,2XA6 ,11H* * *$$$$$)
WAI=WAIC*P21/THETA
IF (IDES.NE.1) GO TO

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PRICF=(PRI0S-1.)/(PRI-1.)
ETAICF=ETAIDS/ETAI
WAICF=WAIDS/WAI
WRITE(6,100)PRICF,ETAICF,WAICF,T21DS
100 FORMAT (23H01.P. COMPRESSOR DESIGN,1X8H PRICF=,E15.8,8H ETAICF=
1E15.8,8H WAICF=,E15.8,8H T21DS=,E15.8)
1 PRI=PRICF*(PRI-1.)+1.
ETAI=ETAICF*ETAI
WAI=WAICF*WAI
CALL THCOMP(PRI,ETAI,T21,H21,S21,P21,T22,H22,S22,P22)
IF(IDES.EQ.1) T22DS=T22
IF(PCBLI.GT.0) BLI=PCBLI*WAI
WA22=WAI-BLI
BLDUZ=PCBL CUI*BLI
BLHPI=PCBL HPI*BLI
BLIPI=PCBL IPI*BLI
BLOBI=PCBL CBI*BLI
PCNI=100.*THETA* CNI/THETA0
CALL COCOMP
RETURN
END

```

```

SUBROUTINE CCOMP
COMMON / ALL/
1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMP ,
2 IGASMX ,IOBURN ,IAFTBN ,IDCO ,IMCO ,IDSHOC ,IMSHOC ,NOZFLT ,
3 IRYS ,LOOPER ,MONAP ,NUMMAP ,MAPECG ,TOLALL ,ERR (9)
COMMON/DESIGN/
1 PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC .
2 ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF .
3 ZIDS ,PCNIQS ,PRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAIQCF .
4 ZCDS ,PCNCCS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF .
5 T4DS ,WF8DS ,DTCCDS ,ETABDS ,WAZCDS ,DPCODS ,DTCCCF ,ETABCF .
6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHFHCF ,T2DS .
7 TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T2IDS .
8 TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS .
9 T24DS ,WF0DS ,DTDUOS ,ETAADS ,WAZ3DS ,OPDUOS ,DTDUCF ,ETAOCF .
A T7DS ,WFAOS ,DTAFOS ,ETAADS ,W66CDS ,DPAFOS ,DTAFCF ,ETAACF .
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 .
CFS55 ,AM55 ,CVNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/FRONT/
1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 .
2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 .
3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 .
4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 .
5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU .
6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB .
7 CNF ,PRI ,ETAI ,WAI ,WAI ,BLOBI ,BLOB ,WA3 .
8 CNC ,PRC ,ETAC ,WACC ,WACC ,ETAB ,DPCOM ,WG4 .
9 CNHP ,ETATHF ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 .
A CNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF .
B CNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS .
C WG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM .
D ALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC .
E WFB ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDU .
F PCBLOUC ,PCCLCBI ,FCBLOBC ,PCBLHPI ,PCBLHPC ,PCBLIFI ,PCBLIPC ,PCBLLP .
G PCBLLFC
COMMON/IPHP/WA22
COMMON / COMP/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),
1 NCN,NPT(15)
DIMENSION WLH(2)
DATA AWORD,WLH/6HCOCMP,6H (LO) ,6H (HI) /
WORD=AWORD
THETA=SQRT (T22/510.688)
IF (IDES.NE.1) GO TO 10
THETA0=THETA
10 CNC=PCNC*THETA0/(100.*THETA)
IF (ZC.LT.0.) ZC=0.
IF (ZC.GT.1.) ZC=1.
CNCS=CNC
CALL SEARCH(ZC,CNC,PRC,WACC,ETAC,
1 CNX(1),NCN,PRX(1,1),WACX(1,1),ETAX(1,1),NPT(1),15,15,IGO)
IF (MODE.EQ.1) GO TO 1
IF ((CNC-CNCS).GT.0.0005*CNC) MAPECG=1
1 IF (IGO.EQ.1.CR.IGO.EQ.2) WRITE(8,1000) CNCS,WLH(IGO)
1000 FORMAT(19H0* * * CNC OFF MAP,F10.4,2XA6,11H* * *$$$$$$)
WAC=WACC*P22/THETA

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IF (IDES.NE.1) GO TO 2
WACDS=WA22
PRCCF=(PRCDS-1.)/(PRC-1.)
ETACCF=ETACDS/ETAC
WACCF=WACDS/WAC
WRITE(6,100)PRCCF,ETACCF,WACCF,T22DS
100  FORMAT(23H0H.P. COMPRESSOR DESIGN,1X8H PRCCF=,E15.8,8H ETACCF=,
1E15.8,8H WACCF=,E15.8,8H T22DS=,E15.8)
2   FRC=PRCCF*(PRC-1.)+1.
    ETAC=ETACCF*ETAC
    WAC=WACCF*WAC
    CALL THCOMP(FRC,ETAC,T22,H22,S22,F22,T3,H3,S3,P3)
    IF(PCBLC.GT.0.) BLC=PCBLC*WAC
    WA3=WAC-BLC
    BLDUC=PCBLDUC*BLC
    BLHPC=PCBLHPC*BLC
    BLIPC=PCBLIPC*BLC
    BLLPC=PCBLIPC*BLC
    BLOB=PCBLOB*BLC
    BLOI=BLDUI+BLDUC
    BLHP=BLHPI+BLHPC
    BLIP=BLIPI+BLIPC
    BLLP=BLLPI+BLLPC
    BLOB=BLDOI+BLOB
    IF(MODE.NE.1) GO TO 3
    IF(ABS(CNC-CNCS).LE.0.001*CNCS) GO TO 4
    WRITE(6,2000)CNCS,CNC
2000  FORMAT(10H0CNC WAS= ,E15.8,11H AND NOW= ,E15.8,
124H CHECK PCNC INPUT$$$$$$)
    CALL ERROR
3   PCNC=100.*THETA*CNC/THETA0
4   ERR(7)=(WAC-WA22)/WAC
    CALL COCOMB
    RETURN
    END

```

SUBROUTINE CCCOMB

COMMON / ALL/

1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IATP ,
 2 IGASMX,ICBURN,IAFTBN,IDCD ,IMCC ,IOSHOC,IMSHOC,NOZFLT,
 3 ITRYS,LCOPEP,NOHAF,NUMMAP,MAPEDG,TOLALL,ERR(9)

COMMON/DESIGN/

1 PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DLMD1 ,DELFG ,DELFN ,DELSFC ,
 2 ZFDS ,PCNFDS ,PRFDS ,ETAZDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
 3 ZIDS ,PCNICS ,PRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
 4 ZCDS ,PCNCCS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
 5 T4DS ,WFBDs ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCCCF ,ETABCF ,
 6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2OS ,
 7 TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21OS ,
 8 TFLPDS ,CNLPDS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22OS ,
 9 T24DS ,WFODS ,CTODS ,ETAODS ,WA23DS ,OPDUDS ,DTDUCF ,ETAOCF ,
 A7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
 BA5 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
 CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,AESAV ,A9SAV ,A28SAV ,A29SAV

COMMON/ FRCNT/

1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
 2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
 3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
 4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
 5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLJU ,
 6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
 7 CNL ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOB ,WA3 ,
 8 CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
 9 CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
 ACNIP ,ETATIP ,DHTCIF ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
 BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
 CAG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
 DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
 EMB ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
 FPCBLDUC,PCBLCBI,PCBLBGC,PCBLHPI,PCBLHPC,PCBLIFI,PCBLIPC,PCBLPI,
 GPCBLIFC

COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),NPS,NPT(15)

DIMENSION G(9),CUMBO(15,15)

DATA AWORD/6HCOCOMB/

WORD=AWORD

G(2)=0.

G(3)=0.

P3PSI=14.696*P3

WA3C=WA3*SGRT(T3)/P3PSI

IF (IDES.EQ.1) WA3CDS=WA3C

DPCOM=DPCODS*(WA3C/WA3CDS)

IF (DPCOM.GT.1.) DPCOM=1.

P4=P3*(1.-DPCOM)

1 IF (T4.GT.3999.) T4=3999.

IF (T4.GE.1000.) GO TO 2

T4=1000.

IF (MODE.EQ.1) MAPEDG=1

2 CTCO=T4-T3

IF (IDES.NE.1) GO TO 3

DTCCCF=DTCCDS/DTCC

3 CTCO=DTCCCF*CTCC

```

F3PSIN=P3PSI
CALL SEARCH(-1.,P3PSIN,DTCO,ETAB,CUMMY,
1PSI(1),NPS,DELT(1,1),ETA(1,1),DUNEO(1,1),NPT(1),15,15,IGO)
IF(IGO.EQ.7) CALL ERROR
4 IF(IDES.NE.1) GO TO 5
ETABCF=ETABDS/ETAB
5 ETAB=ETABCF*ETAB
HV=(((((-.4594317E-19*T4)-.2034116E-15)*T4+.2783643E-11)*T4
1+.2051501E-07)*T4-.2453116E-03)*T4-.9433296E-01)*T4+.1845537E+05
CALL THERMO(P4,HA,T4,XX1,XX2,0,0.0,0)
FAR4=(HA-H3)/(HV*ETAB)
IF(FAR4.LT.0.) FAR4=0.
WFBX=FAR4*WA3
IF(MODE.NE.2) GO TO 8
ERRW=(WFB-WFBX)/WFB
DIR=SQRT(WFB/WFBX)
CALL AFQUIR(Q(1),T4,ERRW,0.,20.,0.0001,DIR,T4T,IGO)
GO TO (6,9,7),IGO
6 T4=T4T
GO TO 1
7 CALL ERROR
8 WFB=WFBX
9 CALL THERMO(F4,H4,T4,S4,XX2,1,FAR4,0)
WG4=WFB*WA3
IF(IDES.EQ.1) WRITE(6,100) WA3COS,ETABCF,DTCCOF
100 FORMAT(17H0COMBUSTOR DESIGN,7X8H WA3COS=,E15.8,8H ETABCF=,E15.8,
18H DTCCOF=,E15.8)
CALL COMPTB
RETURN
END

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SUBROUTINE COMPTB
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX,IOBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOMAP,NUMMAP,MAPEOG,IOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFCS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNICS ,PRIOS ,ETAIDS ,WAIOS ,PRICF ,ETAICF ,WAIKF ,
4ZCDS ,PCNCDS ,PRCOS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WF8DS ,DTGDS ,ETABDS ,WA3CDS ,DPCDS ,DTCCF ,ETABCF ,
6TFHPOS ,CNHPDS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFCDS ,CTDUDS ,ETADDS ,WA23DS ,DPOUDS ,DTDUCF ,ETADCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAI ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIF ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CHG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLGUI ,
FPCBLDUC,PCBLGBI,PCBLOBC,PCBLHFI,PCBLHPC,PCBLIFI,PCBLIPC,PCBLIPI,
GPCBLLFC
COMMON /HTLRB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),
1NTFFS,NPTTFF(15)
DATA AWORD,WLC,WHI/6HCOMPTB,6H(LC),6H(HI) /
WORD=AWORD
IF (IDES.EQ.0) GO TO 1
CNHPCF=CNHPDS*SQRT(T4)/PCNC
1 CNHP=CNHPCF*PCNC/SQRT(T4)
CNHPS=CNHP
TFFHPS=TFFHP
CALL SEARCH(-1.,TFFHP,CNHP,DHTCHP,ETATHP,
1 TFFX(1),NTFFS,CNX(1,1),DHTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,I.C
IF (IGO.EQ. 1.OR.IGO.EQ.11.OR.IGO.EQ.21) WRITE(8,1000)TFFHPS,WLC
IF (IGO.EQ. 2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE(8,1000)TFFHPS,WHI
IF (IGO.EQ.10.OR.IGO.EQ.11.OR.IGO.EQ.12) WRITE(8,2000) CNHPS,WLC
IF (IGO.EQ.20.OR.IGO.EQ.21.OR.IGO.EQ.22) WRITE(8,2000) CNHPS,WHI
1000 FORMAT(19H0****TFFHP OFF MAP,F10.4,2XA6,11H*****$)
2000 FORMAT(19H0**** CNHP OFF MAP,F10.4,2XA6,11H*****$)
IF (IGO.NE.7) GO TO 3
2 CALL ERROR
RETURN

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3  MAPGO=0
   IF (ABS(TFFHPS-TFFHP).LE.0.001*TFFHPS) GO TO 4
   MAPGO=1
   IF (ABS(CNHPS-CNHHP).GT.0.001*CNHPS) MAPGO=3
   GO TO 5
4  IF (ABS(CNHPS-CNHHP).GT.0.001*CNHPS) MAPGO=2
5  IF (MAPGO.GT.0) CALL MAPBAC(1,MAFGC,TFFHPS,TFFHP,CNHPS,CNHHP,PCNC,
1  T4,MODE,NOMAP,NUMMAP)
   IF (NOMAP.GT.0) RETURN
   TFHCAL=MG4*SQRT(T4)/(14.696*P4)
   BTUEXT=0.706705*MPEXT
   DHTCC=(BTUEXT+WAC*(H3-H22))/(MG4*T4)
   IF (IDES.EQ.0) GO TO 6
   TFHPCF=TFHPDS/TFHCAL
   DHHPCF=DHTCC/DHTCHP
   ETHPCF=ETHPDS/ETATHP
   WRITE(6,102) CNHPCF,TFHPCF,ETHPCF,CHHPCF
102 FORMAT(20H0H.P. TURBINE DESIGN,5X7HCNHPCF=,E15.8,8H TFHPCF=,E15.8,
1  8H ETHPCF=,E15.8,8H DHHPCF=,E15.8)
6  TFHCAL=TFHPCF*TFHCAL
   DHTCHP=DHHPCF*DHTCHP
   ETATHP=ETHPCF*ETATHP
   DHTC=DHTCC*T4
   ERR(1)=(TFHCAL-TFFHP)/TFHCAL
   ERR(2)=(DHTCC-DHTCHP)/DHTCC
   CALL THTURB(DHTC,ETATHP,FAR4,H4,S4,P4,T45,H45,S45,P45)
   IF (BLHP.LE.0.) GO TO 7
   FAR45=WFB/(WA3+BLHP)
   MG45=MG4+BLHP
   H45=(BLHPI*H22+BLHPC*H3+MG4*H45)/MG45
   CALL THERMC(P45,H45,T45,S45,XX2,1,FAR45,1)
   GO TO 8
7  FAR45=FAR4
   MG45=MG4
8  CALL COIPTB
   RETURN
   END

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```

SUBROUTINE CCIPTB
COMMON/ ALL/
1WORD ,IDES ,JOES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX ,IOBURN ,IAFTBN ,IDCO ,IMCO ,IOSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOCPER ,NOMAP ,NUMMAP ,MAPEOG ,TOLALL ,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DLMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRI DS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCCS ,PRCDS ,ETA CDS ,WACDS ,PRCCF ,ETA CCF ,WACCF ,
5T4DS ,WF8DS ,DTCODS ,ETABDS ,WA3CDS ,OPCODS ,DTCOCF ,ETABCF ,
6TFHPOS ,CNHPOS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFCDS ,DTCUDS ,ETA DDS ,WA23DS ,DPDUDS ,DTCUCF ,ETA DCF ,
AT7DS ,WFA DS ,DTAFDS ,ETA ADS ,WG6CDS ,OPAFDS ,DTAFCF ,ETA ACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRcnt/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
6CNF ,PRF ,ETA F ,WAF C ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETA I ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETA C ,WACC ,WAC ,ETAB ,OPCOM ,WG4 ,
9CNHP ,ETA HF ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETA TIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETA TLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BL LPC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DAL TP ,ETA R ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFHFP ,TFHIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLOUC ,PCELCBI ,PCBLOBC ,PCBLHPI ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBL LPI ,
GPCBL LPC
COMMON/ITURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),
1NTFFS,NPTTFF(15)
DATA AWORD,WLO,WHI/6HCOIPTB,6H(LC),6H(HI) /
WORD=AWORD
IF(IDES.EQ.0) GO TO 1
CN_PCF=CNIFDS*SQRT(T45)/PCNI
1 CNIP=CNIPCF*PCNI/SQRT(T45)
CNIPS=CNIP
TFFIPS=TFFIP
CALL SEARCH(-1.,TFFIP,CNIP,DHTCIP,ETATIP,
1TFFX(1),NTFFS,CNX(1,1),DHTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,IGO
IF(IGO.EQ.1.OR.IGO.EQ.11.OR.IGO.EQ.21) WRITE(8,1000)TFFIPS,WLC
IF(IGO.EQ.2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE(8,1000)TFFIPS,WHI
IF(IGO.EQ.10.OR.IGO.EQ.11.OR.IGO.EQ.12) WRITE(8,2000)CNIPS,WLO
IF(IGO.EQ.20.OR.IGO.EQ.21.OR.IGO.EQ.22) WRITE(8,2000)CNIPS,WHI
1000 FORMAT(19H0****TFFIP OFF MAP,F10.4,2XA6,11H****$$$$$$)
2000 FORMAT(19H0****CNIP OFF MAP,F10.4,2XA6,11H****$$$$$$)
IF(IGO.NE.7) GO TO 3
2 CALL ERROR
RETURN

```

```

3   MAPGO=0
   IF (ABS(TFFIPS-TFFIP).LE.0.001*TFFIPS) GO TO 4
   MAPGO=1
   IF (ABS(CNIPS-CNIP).GT.0.001*CNIPS) MAPGO=3
   GO TO 5
4   IF (ABS(CNIPS-CNIP).GT.0.001*CNIPS) MAPGO=2
5   IF (MAPGO.GT.0) CALL MAPBAC(3,MAPGC,TFFIPS,TFFIP,CNIPS,CNIP,PCNI,
1T4,MODE,NOMAP,NUMMAP)
   IF (NOMAP.GT.0) RETURN
   TFICAL=MG45*SQRT(T45)/(14.696*P45)
   DHTIC=(MAI*(H22-H21))/(MG45*T45)
   IF (IDES.EG.0) GO TO 6
   TFIPCF=TFIPDS/TFICAL
   DHIPCF=DHTIC/DHTCIP
   ETIPCF=ETIPDS/ETATIP
   WRITE(6,102)CNIPCF,TFIPCF,ETIPCF,DHIPCF
102  FORMAT(20H01.P. TURBINE DESIGN,5X7HCNIPCF=,E15.8,8H TFIPCF=,E15.
10H ETIPCF=,E15.8,8H DHIPCF=,E15.8)
6   TFICAL=TFIPCF*TFICAL
   DHTCIP=DHIFCF*DHTCIP
   ETATIP=ETIFCF*ETATIP
   DHTI=DHTIC*T45
   ERR(8)=(TFICAL-TFFIP)/TFICAL
   ERR(9)=(DHTIC-DHTCIP)/DHTIC
   CALL THTURB (DHTI,ETATIP,FAR45,H45,S45,P45,T5,H5,S5,P5)
   IF (BLIP.LE.0) GO TO 7
   FAR5=WF8/(WA3+BLHP+BLIP)
   WG5=MG45+BLIP
   H5=(BLIPI*H22+BLIPC*H3+MG45*H5)/WG5
   CALL THERMO(P5,H5,T5,S5,XX2,1,FAR5,1)
   GO TO 8
7   FAR5=FAR45
   WG5=MG45
8   CALL COLPT2
   RETURN
   END

```

```

SUBROUTINE COLPTB
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX,IBURN,IAFTBN,ICGD ,IMCD ,IDSHOG,IMSHOG,NGZFLT,
3ITRYS,LOOPER,NDMAP,NUMMAP,HAPEOG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DLMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFOS ,PCNFOS ,PRFOS ,ETAFOS ,NAFOS ,PRFCF ,ETAFCF ,NAFCF ,
3ZIDS ,PCNICS ,PRIDS ,ETAIDS ,NAIDS ,PRICF ,ETAICF ,NAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,NACDS ,PRCCF ,ETACCF ,NACCF ,
5T4OS ,WFOS ,OTCDS ,ETABOS ,HA3CDS ,OPCDS ,OTCCCF ,ETA3CF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,OHHPCF ,T2OS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,OHIPCF ,T21OS ,
8TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,OHLPCF ,T22OS ,
9T24OS ,WFOS ,OTDUS ,ETAADS ,HA23OS ,DPDUS ,OTDUCF ,ETAACF ,
AT7OS ,WFAOS ,OTAFOS ,ETAADS ,HG6CDS ,DPAFOS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNGZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
6CNF ,PRF ,ETAF ,NAFC ,NAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,NAIC ,NAI ,BLOGI ,BLOBC ,HA3 ,
8CNC ,PRC ,ETAC ,NACC ,NAC ,ETAB ,OPCOM ,HG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CNG45 ,FAR45 ,HG5 ,FARE ,HG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EWF8 ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLG ,PCBLDUI ,
FPCBLDUC,PCBLOBI,PCBLOBC,PCBLHPI,PCBLHPC,PCBLIPI,PCBLIPC,PCBLLPI,
GFCBLLPC
COMMON /LTURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),
1NTFFS,NPTTFF(15)
DATA AWORD,WLO,WHI/6HCOLPTB,6H(LC) ,6H(HI) /
WORD=AWORD
IF (IDES.EQ.0) GO TO 1
CNLPCF=CNLPDS*SQRT(T5)/PCNF
1 CNLP=CNLPCF*PCNF/SQRT(T5)
CNLPS=CNLP
TFFLPS=TFFLP
CALL SEARCH(-1.,TFFLP,CNLP,DHTCLP,ETATLP,
1TFFX(1),NTFFS,CNX(1,1),DHTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,IGO)
IF (IGO.EQ. 1.CR.IGO.EQ.11.OR.IGO.EQ.21) WRITE(8,1000)TFFLPS,WLO
IF (IGO.EQ. 2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE(8,1000)TFFLPS,WHI
IF (IGO.EQ. 10.OR.IGO.EQ.11.OR.IGO.EQ.12) WRITE(8,2000) CNLPS,WLO
IF (IGO.EQ. 20.OR.IGO.EQ.21.OR.IGO.EQ.22) WRITE(8,2000) CNLPS,WHI
1000 FORMAT(19H0****TFFLP OFF MAP,F10.4,2XA6,11H*****$$$$$$)
2000 FORMAT(19H0**** CNLP OFF MAP,F10.4,2XA6,11H*****$$$$$$)
IF (IGO.NE.7) GO TO 3
2 CALL ERROR
RETURN

```

```

3     MAPGO=0
      IF (ABS(TFFLPS-TFFLP).LE.0.001*TFFLPS) GO TO 4
      MAPGO=1
      IF (ABS(CNLPS-CNLP).GT.0.001*CNLPS) MAPGO=3
      GO TO 5
4     IF (ABS(CNLPS+CNLP).GT.0.001*CNLPS) MAPGO=2
5     IF (MAPGO.GT.0) CALL MAPBAC(2,MAPGO,TFFLPS,TFFLP,CNLPS,CNLP,PCNF,
1  T4,MODE,NOMAP,NUMMAP)
      IF (NOMAP.GT.0) RETURN
      TFLCAL=NG5*SQRT(T5)/(14.696*P5)
      DHTCF=MAF*(H21-H2)/(NG5*T5)
      IF (IDES.EQ.0) GO TO 6
      TFLPCF=TFLPDS/TFLCAL
      CHLPCF=DHTCF/DHTCLP
      ETLPCL=ETLPDS/ETATLP
102  WRITE(6,102)CNLPCF,TFLPCF,ETLPCL,CHLPCF
      FORMAT(20HDL.P. TURBINE DESIGN,5X7HCNLPCF=,E15.8,8H TFLPCF=,E15.8
1  8H ETLPCL=,E15.8,8H DHLPCF=,E15.8)
6     TFLCAL=TFLPCF*TFLCAL
      DHTCLP=DHLPCF*DHTCLP
      ETATLP=ETLPCF*ETATLP
      DHTF=DHTCF*T5
      ERR(3)=(TFLCAL-TFFLP)/TFLCAL
      ERR(4)=(DHTCF-DHTCLP)/DHTCF
      CALL THTURB(DHTF,ETATLP,FAR5,H5,S5,P5,T55,H55,S55,P55)
      IF (BLLP.LE.0.) GO TO 7
      FAR55=MF8/(MA3+R.LHP+BLLP+BLLP)
      NG55=NG5+BLLP
      H55=(BLLP1*H22+BLLPC*H3+NG5*H55)/NG55
      CALL THERMO(F55,H55,T55,S55,XX2,1,FAR55,1)
      GO TO 8
7     FAR55=FAR5
      NG55=NG5
8     CALL FRTOSC
      RETURN
      END

```

SUBROUTINE FRTOSD
COMMON/ FRONT/

```

1 T1      , F1      , H1      , S1      , T2      , P2      , H2      , S2      ,
2 T21     , F21     , H21     , S21     , T22     , P22     , H22     , S22     ,
3 T3      , P3      , H3      , S3      , T4      , P4      , H4      , S4      ,
4 T45     , P45     , H45     , S45     , T5      , P5      , H5      , S5      ,
5 T55     , P55     , H55     , S55     , BLF     , BLI     , BLC     , BLOU    ,
6 CNF     , PRF     , E1AF    , WAF     , WAF     , BLDUI   , BLDUC   , BLOB    ,
7 CNX     , PRI     , ETAI    , WAIC    , WAI     , BLOBI   , BLOBC   , WA3     ,
8 CNC     , PRC     , ETAC    , WACC    , WAC     , ETAB    , OPCOM   , W4      ,
9 CNHP    , ETATHP   , DHTCHP  , DHTC    , BLHP    , BLHPI   , BLHPC   , FAR4    ,
ACNIP     , ETATIP   , DHTCIP  , DHTI    , BLIP    , BLIPI   , BLIPC   , DUMF    ,
BCNLP     , ETATLP   , DHTCLP  , DHTF    , BLLP    , BLLPI   , BLLPC   , CS      ,
CNG45    , FAR45    , W45     , FAR5    , W55     , FAR55   , HPEXT   , AM      ,
DALT     , ETAR     , ZF      , PCNF    , ZI      , PCNI    , ZC      , PCNC    ,
EWF      , TFFIP    , TFFIP    , TFFLP   , PCBLF   , PCBLI   , PCBLC   , PCBLOUI ,
FFCBLOUC , PCBLOBI , PCBLOC  , PCBLPI  , PCBHPC  , PCBLIPI , PCBIPC  , PCBLLPI ,
GPCBLIPC

```

COMMON/ SIDE/

```

1 XP1     , XWAF    , XWAI    , XWAC    , XBLF    , XBLOU   , XBLOUI  , XBLOUC  ,
2 XH22    , XH3     , XT21    , XP21    , XH21    , XS21    , DUMS1   , DUMS2   ,
3 T23     , P23     , H23     , S23     , T24     , P24     , H24     , S24     ,
4 T25     , P25     , H25     , S25     , T26     , P26     , H26     , S26     ,
5 T29     , P29     , H29     , S29     , DUMS3   , DUMS4   , DUMS5   , DUMS6   ,
6 W40     , WFL     , W424    , FAR24   , ET40    , DPOUC   , BYPASS  , DUMS7   ,
7 TS29    , PS28    , V28     , AM28    , TS29    , PS29    , V29     , AM29

```

XP1=P1

XWAF=WAF

XWAI=WAI

XWAC=WAC

XBLF=ELF

XBLOU=BLOU

XBLOUI=BLOUI

XBLOUC=BLOUC

XH22=H22

XH3=H3

XT21=T21

XP21=P21

XH21=H21

XS21=S21

CALL CODUCT

RETURN

END

```

SUBROUTINE CODUCT
COMMON / ALL/
1 WORD ,IDES ,JOES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2 IGASMX,IOBURN,IAFTBN,IOCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3 ITRYS,LOOPER,NOMAP,NUMMAP,MAPEDG,TOLALL,ERR(9)
COMMON/DESIGN/
1 PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,GUMD1 ,DELFG ,DELFN ,DELSFC ,
2 ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3 ZIDS ,PCNIGS ,PRIOS ,ETAIDS ,WAIOS ,PRICF ,ETAICF ,WAI CF ,
4 ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5 T4DS ,WFEDS ,DTGDS ,ETABDS ,WA3CDS ,OPCODS ,DTCCOF ,ETABCF ,
6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T20S ,
7 TFIPDS ,CNIFDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIFCF ,T210S ,
8 TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T220S ,
9 T24DS ,WFCD S ,DTDUDS ,ETAADS ,WA23DS ,OPDUDS ,DTDUCF ,ETAACF ,
AT7DS ,WFA DS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A9SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ SIDE/
1 P1 , WAF , WAI , WAC , ELF , BLDU , BLDUI , BLDUC ,
2 H22 , H3 , T21 , P21 , H21 , S21 , DUMS1 , DUMS2 ,
3 T23 , P23 , H23 , S23 , T24 , P24 , H24 , S24 ,
4 T25 , P25 , H25 , S25 , T28 , P28 , H28 , S28 ,
5 T29 , P29 , H29 , S29 , DUMS3 , DUMS4 , DUMS5 , DUMS6 ,
6 WAD , WFC , WG24 , FAR24 , ETAD , DPOUC , BYPASS , DUMS7 ,
7 TS28 , PS28 , V28 , AM28 , TS29 , PS29 , V29 , AM29
DIMENSION Q(9)
DATA ANORD1,ANORD2/6HCODUCT,6HDNOZZL/
WORD=ANORD1
Q(2)=0.
Q(3)=0.
WAX=WAF-WAI-ELF
WAD=WAX+BLDU
P23=P21
H23=(BLDUI*H22+BLDUC*H3+WAX*H21)/WAD
CALL THERMC(P23,H23,T23,S23,XX2,1,0.0,1)
BYPASS=(WAF-WAI)/WAI
WA23C=WAD*SORT(T23)/P23
IF (IDES.EQ.1) WA23DS=WA23C
OPDUC=OPDUDS*(WA23C/WA23DS)
IF (OPDUC.GT.1.) OPDUC=1.
P24=P23*(1.-OPDUC)
IF (IGASMX.GT.0) IOBURN=0
IF (IOBURN.NE.0) GO TO 2
T24=T23
WFD=0.
FAR24=0.
GO TO 7
2 IF (IOBURN.EQ.2) T24=T23+2000.
3 IF (T24.GT.4000.) T24=4000.
IF (T24.LT.T23) T24=T23
C *** IF DESIRED, ENTER CALCULATIONS FOR ETAD HERE
HV=((((-4594317E-19*T24)-.2034116E-15)*T24+.2783643E-11)*T24+
1.2051501E-07)*T24-.2453116E-03)*T24-.9433296E-01)*T24+.1845537E+
CALL THERMC(P24,WA,T24,XX1,XX2,0,0.0,0)

```



```

FAR24=(HA-H23)/(HV*ETA0)
IF (FAR24.LT.0.) FAR24=0.
WFOX=FAR24*WAD
IF (IDBURN.NE.2) GO TO 6
ERRW=(WFO-WFOX)/WFO
DIR=SQRT(WFO/WFCX)
CALL AFQUIR(Q(1),T24,ERRW,0.,29.,0.0001,DIR,T24T,IGO)
GO TO (4,7,5),IGO
4 T24=T24T
GO TO 3
5 CALL ERROR
6 WFO=WFOX
7 CALL THERMO(P24,H24,T24,S24,XX2,1,FAR24,0)
WG24=WFO+WAD
IF (IDES.EQ.1) WRITE(6,101) WA23DS
101 FORMAT(12HDUCT DESIGN,12X8H WA23DS=,E15.8)
*** IF DESIRED, ENTER OTHER LOSSES HERE
T25=T24
P25=P24
H25=H24
S25=S24
IF (IGASM.GT.0) GO TO 11
WORD=AWORD2
A28SAV=A28
A29SAV=A29
NOZD=0
IDNOZ=0
IF (NOZFLT.EQ.2.OR.NOZFLT.EQ.3) NOZD=1
IF (IDES.EQ.1.OR.IDBURN.GT.0.OR.NOZD.EQ.1) IDNOZ=1
IF (IDCD.EQ.1) GO TO 8
CALL CONVRG(T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28, P25R,
1 T28,H28,P28,S28,TS28,PS28,V28,AM28,ICON)
GO TO (9,9,3,5),ICON
CALL CONDV(T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28,A29,P25R,
1 T28,H28,P28,S28,T29,H29,P29,S29,TS28,TS29,PS28,PS29,V28,V29,A
2 AM29,ICON)
IDSHOC=ICON
GO TO (10,10,10,5),ICGN
3 T29=T28
H29=H28
P29=P28
S29=S28
TS29=TS28
PS29=PS28
V29=V28
AM29=AM28
A29=A28
IDSHOC=ICON+3
10 ERR(5)=(P25R-P25)/P25R
IF (IDNOZ.EQ.1) WRITE(6,100) A28,AM28,A29,AM29
100 FORMAT(19HDUCT NOZZLE DESIGN,5X8H A28=,E15.8,8H AM28=,E1
18H A29=,E15.8,8H AM29=,E15.8)
11 CALL FASTBK
RETURN
END

```

SUBROUTINE FASTBK

COMMON/ FRCNT/

```

1T1      ,P1      ,H1      ,S1      ,T2      ,P2      ,H2      ,S2      ,
2T21     ,P21     ,H21     ,S21     ,T22     ,P22     ,H22     ,S22     ,
3T3      ,P3      ,H3      ,S3      ,T4      ,P4      ,H4      ,S4      ,
4T45     ,P45     ,H45     ,S45     ,T5      ,P5      ,H5      ,S5      ,
5T55     ,P55     ,H55     ,S55     ,BLF     ,BLI     ,BLC     ,BLOU     ,
6CNF     ,PRF     ,ETAF    ,WAF     ,WAF     ,BLDUI   ,BLOUC   ,BLOB     ,
7CNI     ,PRI     ,ETAI    ,WAI     ,WAI     ,BLOBI   ,BLOBC   ,WA3      ,
8CNC     ,PRC     ,ETAC    ,WACC    ,WAC     ,ETAB    ,OPCOM   ,WG4      ,
9GNHP    ,ETATHP ,DHTCHP ,DHTC    ,BLHP    ,BLHPI   ,BLHPC   ,FAR4     ,
ACNIP    ,ETATIP ,DHTCIP ,DHTI    ,BLIP    ,BLIPI   ,BLIPC   ,DUMF     ,
BCNLP    ,ETATLP ,DHTCLP ,DHTF    ,BLLP    ,BLLPI   ,BLLPC   ,CS       ,
CHG45    ,FAR45 ,WG5     ,FAR5    ,WG55    ,FAR55   ,HPEXT   ,AM       ,
DALTP    ,ETAR    ,ZF      ,PCNF    ,ZI      ,PCNI    ,ZC      ,PCNC     ,
EWF8     ,TFFHF   ,TFFIP   ,TFFLP   ,PCBLF   ,PCBLI   ,PCBLC   ,PCBLDUI ,
FPCBLOUC,PCBLOBI,PCBLOBC,PCBLHPI,PCBLHPC,PCBLIPI,PCBLIPC,PCBLLPI,
GPCBLFC

```

COMMON/ SIDE/

```

1XP1     ,XWAF    ,XWAI    ,XWAC    ,XBLF    ,XBLOU   ,XBLOUI   ,XBLOUC   ,
2XH22    ,XH3      ,XT21    ,XP21    ,XH21    ,XS21    ,DUMS1    ,DUMS2    ,
3T23     ,P23      ,H23     ,S23     ,T24     ,P24     ,H24     ,S24     ,
4T25     ,P25      ,H25     ,S25     ,T28     ,P28     ,H28     ,S28     ,
5T29     ,P29      ,H29     ,S29     ,DUMS3   ,DUMS4   ,DUMS5   ,DUMS6   ,
6WAD     ,WFD      ,WG24    ,FAR24   ,ETAD    ,DPOUC   ,BYPASS   ,DUMS7   ,
7TS28    ,PS28     ,V28     ,AM28    ,TS29    ,PS29    ,V29     ,AM29

```

COMMON / EACK/

```

XXT55    ,XP55    ,XH55    ,XS55    ,XT25    ,XP25    ,XH25    ,XS25    ,
XXWF8    ,XWG55   ,XFAR55  ,XWFD    ,XWG24   ,XFAR24  ,XXP1    ,DUMB     ,
3T6      ,P6      ,H6      ,S6      ,T7      ,P7      ,H7      ,S7      ,
4T8      ,P8      ,H8      ,S8      ,T9      ,P9      ,H9      ,S9      ,
5WG6     ,WFA     ,WG7     ,FAR7    ,ETAA    ,OPAFI   ,V55     ,V25     ,
6PS6     ,V6      ,AM6     ,TS7     ,PS7     ,V7      ,AM7     ,AM25    ,
7TS8     ,PS8     ,V8      ,AM8     ,TS9     ,PS9     ,V9      ,AM9     ,
8VA      ,FRD     ,VJD     ,FGMD    ,VJM     ,FGMM    ,FGPD    ,FGPM    ,
9FGM     ,FGP     ,WFT     ,WGT     ,FART    ,FG      ,FN      ,SFC

```

XT55=T55

XP55=P55

XH55=H55

XS55=S55

XT25=T25

XP25=P25

XH25=H25

XS25=S25

XWF8=WFB

XWG55=WG55

XFAR55=FAR55

XWFD=WFD

XWG24=WG24

XFAR24=FAR24

XXP1=P1

CALL COMIX

RETURN

END

```

SUBROUTINE COMIX
DIMENSION QQ(9)
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX, IDBURN, IAFTBN, IDCD ,IMCD ,IDSHOC, IMSHOC, NOZFLT,
3ITRYS, LOOPER, NOMAP, NUMMAP, MAPEDG, TOLALL, ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIDS ,ETAIDS ,WAFDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WF8DS ,DTCODS ,ETA8DS ,WA3CDS ,OPCODS ,DTCCGF ,ETABCF ,
6TFHPOS ,CNHPOS ,EYHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPOS ,CNIPOS ,ETIPOS ,TFIPCF ,CNI PCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPOS ,CNLPOS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFCDS ,DTUUDS ,ETA0DS ,WA23DS ,DPUUDS ,DTUUCF ,ETA0CF ,
AT7DS ,WFA0S ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/FRONT/DUMF1(98),ZF,PCNF,DUMF2(21)
COMMON / EACK/
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2WF8 ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WG7 ,FART ,FG ,FN ,SFC
DATA AMORD/6H COMIX/
WORD=AMORD
AJ=778.26
CAPSF=2116.2170
G=32.174049
CALL PROCOM(FAR55,T55,XX1,XX2,XX3,XX4,PHI55,XX5)
CALL PROCOM(FAR24,T25,XX1,XX2,XX3,XX4,PHI25,XX5)
IF (IDES.EQ.0) GO TO 6
C *** CALCULATE A55 AND A25 WITH PS25=PS55
IF (PS55.EQ.0.) GO TO 50
TS55=TS55*(PS55/P55)**0.286
DO 1 I=1,15
CALL PROCOM(FAR55,TS55,CS55,AK55,CP55,REX55,PHI55,HS55)
PHIS=PHI55-REX55*ALOG(P55/PS55)
DELPHI=PHIS-PHI55
IF (ABS(DELPHI).LE.0.0001*PHIS) GO TO 3
1 TS55=TS55*EXP(4.0*DELPHI)
2 CALL ERROR
RETURN
50 TS55=0.875*T55
DO 51 I=1,15
CALL PROCOM(FAR55,TS55,CS55,AK55,CP55,REX55,PHI55,HS55)
V55=AM55*CS55
HSCAL=H55-V55**2/(2.*G*AJ)
DELHS=HSCAL-HS55

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```

51 IF (ABS(DELHS) .LE. 0.0055 * HSCAL) GO TO 52
   TS55 = TS55 + DELHS / CP55
   GO TO 2
52 P55 = P55 / EXP ((PHI55 - PHIS55) / REX55)
   3 IF (H55 .GT. HS55) GO TO 53
   WRITE (8, 101) P55, PS55, T55, TS55, H55, HS55
101 FORMAT (22H0SQRT OF H55-HS55 NEG ,6E15.6,6H$$$$$$)
   CALL ERROR
53 V55 = SQRT (2. * G * AJ * (H55 - HS55))
   RHO = CAPSF * PS55 / (AJ * REX55 * TS55)
   A55 = WG55 / (RHO * V55)
   AM55 = V55 / CS55
   IF (IGASHX .GT. 0) GO TO 54
   WRITE (6, 104) A55, AM55
104 FORMAT (20HOTURBINE AREA DESIGN, 6X6H A55=,E15.8,8H AM55=,E15.8
   GO TO 34
54 PS25 = PS55
   TS25 = T25 * (PS25 / P25) ** 0.286
   DO 4 I=1,15
   CALL FROCOM (FAR24, TS25, CS25, AK25, CP25, REX25, PHIS25, HS25)
   PHIS = PHI25 - REX25 * ALOG (P25 / PS25)
   DELPHI = PHIS - PHIS25
   IF (ABS(DELPHI) .LE. 0.0001 * PHIS) GO TO 5
   4 TS25 = TS25 * EXP (4.0 * DELPHI)
   GO TO 2
   5 IF (H25 .GT. HS25) GO TO 55
   WRITE (8, 102) P25, PS25, T25, TS25, H25, HS25
102 FORMAT (22H0SQRT OF H25-HS25 NEG ,6E15.6,6H$$$$$$)
   CALL ERROR
55 V25 = SQRT (2. * G * AJ * (H25 - HS25))
   RHO = CAPSF * PS25 / (AJ * REX25 * TS25)
   A25 = WG24 / (RHO * V25)
   AM25 = V25 / CS25
   WRITE (6, 100) A55, AM55, A25, AM25
100 FORMAT (25HOTURBINE/DUCT AREA DESIGN, 7H A55=,E15.8,
   18H AM55=,E15.8,8H A25=,E15.8,8H AM25=,E15.8)
   GO TO 20
C *** CALCULATE PS55 AND PS25
6 WQA = WG55 / A55
   C1 = P55 * SQRT (G / (T55 * AJ)) * CAPSF
   MCON = 0
   QD (2) = 0.
   QD (3) = 0.
   AM55 = 0.50
   TS55 = 0.875 * T55
7 DO 8 I=1,15
   CALL FROCOM (FAR55, TS55, CS55, AK55, CP55, REX55, PHIS55, HS55)
   V55 = AM55 * CS55
   HSCAL = H55 - V55 ** 2 / (2. * G * AJ)
   DELHS = HSCAL - HS55
   IF (ABS(DELHS) .LE. 0.0005 * HSCAL) GO TO 9
   8 TS55 = TS55 + DELHS / CP55
   GO TO 2
   9 WQAT = C1 * SQRT (AK55 / REX55) * AM55 / (1. + (AK55 - 1.) * AM55 ** 2 / 2.) **
   1 ((AK55 + 1.) / (2. * (AK55 - 1.)))

```

```

AMX=AM55
IGOGO=0
10 DIR=WQA/WQAT
EH=(WQA-WQAT)/WQA
CALL AFQUIR(QQ(1),AMX,EH,0.,30.,0.0005,DIR,AMXT,ICON)
GO TO (11,15,2),ICON
11 IF(AMXT.LE.1.0) GO TO 13
AMXT=0.7
MCON=MCON+1
IF(MCON.LE.1) GO TO 13
IF(MOCE.EQ.3) GO TO 120
WRITE(8,103)FCNF,AMX,P55,PS55,P25,PS25
103 FORMAT(12HOCOMIX PCNF=,F7.4,4H AM=,F8.6,5H P55=,F9.5,
16H PS55=,F9.5,5H P25=,F9.5,6H PS25=,F9.5,(H$$$$$$)
PCNF=1.01*PCNF
12 NOMAP=7
RETURN
120 WRITE(8,121)ZF,AMX,P55,PS55,P25,PS25
121 FORMAT(10HOCCHIX ZF=,F8.5,4H AM=,F8.6,5H P55=,F9.5,
16H PS55=,F9.5,5H P25=,F9.5,6H PS25=,F9.5,6H$$$$$$)
ZF=0.99*ZF
GO TO 12
13 IF(IGOGO.EQ.1) GO TO 14
AM55=AMXT
GO TO 7
14 AM25=AMXT
GO TO 16
15 IF(IGOGO.EQ.1) GO TO 19
PS55=P55/EXP((PHI55-PHIS55)/REX55)
IF(IGASHX.LE.0) GO TO 34
WQA=WG24/A25
C1=P25*SQRT(G/(T25*AJ))*CAPSF
MCON=0
QQ(2)=0.
QQ(3)=0.
AM25=0.25
TS25=.875*T25
16 DO 17 I=1,15
CALL PROCOM(FAR24,TS25,CS25,AK25,CP25,REX25,PHIS25,HS25)
V25=AM25*CS25
HSCAL=H25-V25**2/(2.*G*AJ)
DELHS=HSCAL-HS25
IF(ABS(DELHS).LE.0.0005*HSCAL) GO TO 18
17 TS25=TS25+DELHS/CP25
GO TO 2
18 WQAT=C1*SQRT(AK25/REX25)*AM25/(1.+(AK25-1.)*AM25**2/2.)*
1((AK25+1.)/(2.*(AK25-1.)))
AMX=AM25
IGOGO=1
GO TO 10
19 PS25=P25/EXP((PHI25-PHIS25)/REX25)
2) WG6=WG24+WG55
ERR(5)=(PS25-PS55)/PS25
WF6=WFD+WF0
FAR6=WF6/(WG6-WF6)

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H6=(.HG24*H25+.HG55*H55)/HG5
CALL THERMO(1.,H5,T6,PHI6,AM6,1,FARL,1)
C1=PS55*A55*(1.+AK55*AM55**2)+PS25*A25*(1.+AK25*AM25**
TS6=0.833*TE
DO 25 I=1,15
CALL PROCOP(FAR6,TS6,CS6,AK6,CP6,REX6,PHIS6,HS6)
C2=HG6*SQRT(AJ*REX6*TE/(AK6*U))
C3=C2/(GAPSF*C1)
C4=(AK6-1.)/2.-(C3*AK6)**2
C5=1.-2.*AK6*C3**2
C6=C5**2+4.*C4*C3**2
IF (C6)21,22,23
21 CALL ERROR
RETURN
22 AM62G=-C5/(2.*C4)
GO TO 24
23 AM62G=(SQRT(C6)-C5)/(2.*C4)
24 IF (AM62G.LE.0.) GO TO 21
AM6G=SQRT(AM62G)
V6=AM6G*CS6
HSCAL=H6-V6**2/(2.*G*AJ)
DELHS=HSCAL-H56
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 26
2: TS6=TS6+DELHS/CF6
GO TO 21
26 IF (IGASHX.CT.0) A6G=A25+A55
C7=SQRT(1.+(AK6-1.)*AM62G/2.)
FS6=C2/(GAPSF*A6G*AM6G*C7)
F6=PS6*EXP((PHI6-PHIS6)/REX6)
CALL THERMO(P6,H6,T6,S6,XX1,1,FAR6,G)
S6AVE=(.HG24*S25+HG55*S55)/HG6
IF (S6.GE.S6AVE) GO TO 27
S6=S6AVE
P6=EXP(AMX*(PHI6-S6)/1.986375)
27 IF (IGASNX.EC.1) GO TO 35
IF (IDES.EC.0) GO TO 35
C *** CALCULATE A6 AS A FUNCTION OF INPUT AM6
TS6P=TE/(1.0+(((AK6-1.0)/2.0)*AM6**2.))
DO 28 JJ=1,15
CALL PROCOP(FAR6,TS6P,CS6,AK6,CP6,REX6,PHIS6,HS6)
V6=AM6*CS6
HSCAL=H6-V6**2/(2.*G*AJ)
DELHS=HSCAL-H56
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 2
28 TS6=TS6P+DELHS/CF6
GO TO 28
29 FS6=F6/(1.0+(((AK6-1.0)/2.0)*AK6**2.))**((AK6/(AK6-1.0)).
AM6DS=AM6
RHO=GAPSF*FS6/(AJ*REX6*TS6)
AG=HG6/(RHO*V6)
WRITE(6,230) A6
200 FORMAT(1H6, 31AFTER:JUNNER ENTRANCE DESIGN AREA A6 ,F8.3)
GO TO 36
C *** CALCULATES AM6=F(A6DESIGN)
30 TS6P=TE/(1.0+(((AK6-1.0)/2.0)*AM6CS**2.))

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DO 32 I=1,15
CALL PROCOM (FAR6,TS6P,CS6,AK6,CPE,REX6,PHIS6,HS6)
PS6P=PS6*(TS6P/TS6)**(AK6/(AK6-1.0))
RHO=CAPSF*FS6P/(AJ*REX6*TS6P)
IF (H6.GT.HS6) GO TO 31
WRITE (8,201)P6,PS6P,T6,TS6P
201  FORMAT(20H0SORT OF H6-HS6 NEG ,6E15.6,6H$$$$$$)
CALL ERROR
31  V6=SQRT(2.*G*AJ*(H6-HS6))
    A6P=HG6/(RHO*V6)
    DELA6=A6P-A6
    V6=HG6/(RHO*A6)
    AM6=V6/CS6
    AM62=AM6**2.
    IF (ABS(DELA6).LE.0.002*AF) GO TO 33
32  TS6P=T6/(1.0+(((AK6-1.0)/2.0)*AM62))
    GO TO 21
33  TS6=TS6P
    PS6=PS6P
    GO TO 36
34  T6=T55
    P6=P55
    H6=H55
    S6=S55
    HG6=HG55
    PS6=PS55
    V6=V55
    AM6=AM55
    IF (IGASHX.EQ.0) A6=A55
    GO TO 36
35  AM62=AM62G
    AM6=AM6G
    A6=A25+A55
36  CALL COAFBN
    RETURN
    END

```

```

SUBROUTINE COAFBN
COMMON / ALL/
1WORD ,IDES ,JOES ,KDES ,MODE ,INIT ,IDUMP ,IAHTP ,
2IGASHX,ICBURN,IAFTBN,ICDN ,IMCO ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOHAP,NUMHAP,MAPEDG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF
3ZIOS ,PCNICS ,PRI. ) ,ETAIOS ,WAIDS ,PRICF ,ETAICF ,WAICF
4ZCOS ,PCNCOS ,PRCOS ,ETACOS ,WACOS ,PRCCF ,ETACCF ,WACCF
5T4DS ,WF8DS ,OTCDS ,ETABOS ,WA3COS ,OPCOS ,DTCOCF ,ETA3CF
6TFHPDS ,CNHPDS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS
7TFIPDS ,CNIPCS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS
8TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS
9T4DS ,WFDS ,OTDUDS ,ETAADS ,WA23DS ,OPDUDS ,DTDUCF ,ETAADC
AT7DS ,WAFDS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A26 ,A29
CPS55 ,AM55 ,CVDNOZ ,CVHNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON / BACK/
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6FS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMH ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC

```

DIMENSION C(9)

DATA AWORD/6HCOAFBN/

WORD=AWORD

C(2)=0.

Q(3)=0.

AJ=778.26

CAPSF=2116.2170

G=32.174049

WF6=WFB

IF (IGASHX.GT.0) WF6=WF6+WFD

WA6=WG6-WF6

C *** CRY LOSS

WG6C=WG6*SCR1(T6)/P6

2 IF (IDES.EQ.1) WG6CDS=WG6C

DPAFT=DPAFDS*(WG6C/WG6CDS)

IF (DPAFT.GT.1.) DPAFT=1.

P7=P6*(1.-DPAFT)

A7=A6

FAR6=WFB/WAC

CALL PROCOM(FAR6,T6,XX1,XX2,XX3,XX4,PHI6,XX6)

WQA=WG6/A7

C1=P7*SQRT(G/(T6*AJ))*CAPSF

AM7=AM6

TS7=0.875*T6

20 DO 22 I=1,15

CALL PROCOM(FAR6,TS7,CS7,AK7,CP7,REX7,PHI7,KS7)

V7=AM7*CS7


```

HSCAL=H6-V7**2/(2.*G*AJ)
DELHS=HSCAL-HS7
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 24
22 TS7=TS7+DELHS/CF7
GO TO 8
24 WQAT=C1*SQRT(AK7/REX7)*AM7/(1.+(AK7-1.)*AM7**2/2.)**
1 ((AK7+1.)/(2.*(AK7-1.)));
DIR=WQA/WQAT
EW=(WQA-WQAT)/WQA
CALL AFQUIR(Q(1),AM7,EW,0.,30.,0.0005,DIR,AM7T,IGO)
GO TO (26,28,8),IGO
26 AM7=AM7T
IF (AM7.GE.1.0) AM7=0.9
GO TO 20
28 PS7=P7/EXP((PHI6-PHI7)/REX7)
IF (IAFTBN.GT.0) GO TO 4
C *** NON-AFTERBURNING
3 T7=T6
WFA=0.0
FAR7=FAR6
WG7=WG6
GO TO 13
C *** AFTREBURNING
4 IF (IAFTBN.EQ.2) T7=T6+2000.
IF (T7.LE.T6) GO TO 3
RHO65=CAPSF*PS7/(AJ*REX7*TS7)
PS65=PS7
V65=V7
Q(2)=0.
Q(3)=0.
5 IF (T7.GT.4000.) T7=4000.
C *** IF DESIRED, ENTER CALCULATIONS FOR ETAA HERE
HV=((((( (-.4594317E-19*T7)-.2034116E-15)*T7+.2783643E-11)*T7
1+.2051501E-07)*T7-.2453116E-03)*T7-.9433296E-01)*T7+.1845537E+05
CALL THERMO(F7,HA,T7,XX1,XX2,0,0.0,0)
FAR7=(HA-H6)/(HV*ETAA)
IF (FAR7.GT.0.) GO TO 6
T7=T6
GO TO 5
6 WFAX=FAR7*WG6
IF (IAFTBN.EQ.1) GO TO 9
ERRW=(WFA-WFAX)/WFA
DIR=SQRT(WFA/WFAX)
CALL AFQUIR(Q(1),T7,ERRW,0.,20.,0.0001,DIR,T7T,IGO)
GO TO (7,10,8),IGO
7 T7=T7T
GO TO 5
8 CALL ERROR
9 WFA=WFAX
10 FAR7=(WF6+WFA)/WA6
WG7=WG6+WFA
C *** MOMENTUM LCSS
CALL PROCOP(FAR7,T7,XX1,XX2,XX3,REX7,PHI7,H7)
RHO7=CAPSF*P7/(AJ*REX7*T7)
V7=WG7/(RHC7*A7)

```

```

Q(2)=0.
Q(3)=0.
FS7=PS65-0.01
11  RH07=HG7/(V7*A7)
    HS7=H7-V7**2/(2.*G*AJ)
    CALL THERMO(1.0,HS7,TS7,PHIS7,XX2,1,FAR7,1)
    IF(TS7.GE.301.) GO TO 110
    CALL THERMC(1.0,HS7,400.,PHIS7,XX2,1,FAR7,0)
    V7=SQRT(2.*G*AJ*(H7-HS7))
    GO TO 11
110  PS7=RK07*AJ*REX7*TS7/CAPSF
    PS7A=PS65+(RH065*V65**2-RH07*V7**2)/(L    °SF)
    DIR=SQRT(ABS(PS7/PS7A))
    EP=(PS7-PS7A)/PS7
    CALL AFQUIR(Q(1),V7,EP,0.,50.,0.0005,DIR,V7T,IGO)
    V7=V7T
    IF(V7.LT.100.) V7=100.
    GO TO (11,12,8),IGO
12  P7=PS7*EXP((PHI7-PHIS7)/REX7)
    CALL FROCOM(FAR7,TS7,CS7,XX2,XX3,XX4,XX5,XX6)
    AM7=V7/CS7
13  CALL THERMO(P7,H7,T7,S7,XX2,1,FAR7,0)
    IF(IDES.EQ.1) WRITE(6,100) WG6CDS
100  FORMAT(19H0AFTERBURNER DESIGN,5X8H WG6CDS=,E15.8)
    CALL COMNOZ
    RETURN
    END

```

```

SUBROUTINE COMNOZ
COMMON / ALL/
1 WORD ,IDES ,JOES ,KOES ,MODE ,INIT ,IDUMP ,IAMTP ,
2 IGASHY,IOBURN,IAFTBN,IDCD ,IMCD ,IOSHOC,INSHOC,NOZFLT,
3 IT RYS,LOOPER,NOHAP,NUHMAP,MAPEDG,TOLALL,ERR(9)
COMMON/DESIGN/
1 PCNFGU ,PCNIGU ,PCNGGU ,T4GU ,DLMQ1 ,DELFG ,DELFN ,DELSFC ,
2 ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3 ZIOS ,PCNIOS ,PRIOS ,ETAIOS ,WAIOS ,PRICF ,ETAICF ,WAI CF ,
4 ZCOS ,PCNCOS ,PRCOS ,ETACOS ,WACOS ,PRCCF ,ETACCF ,WACCF ,
5 Y4DS ,WFBOS ,DTCOOS ,ETABOS ,WAB3COS ,OPCOOS ,DTCCOF ,ETABCF ,
6 TFHPOS ,GNHPOS ,ETHPOS ,TFHPCF ,GNHPCF ,ETHPCF ,DHHPCF ,T2OS ,
7 TFIPOS ,GNFIPOS ,ETIPOS ,TFIPCF ,GNIPCF ,ETIPCF ,DHIPCF ,T2IOS ,
8 TFLPOS ,GNLPOS ,ETLPOS ,TFLPCF ,GNLPCF ,ETLPCF ,DHLPCF ,T22OS ,
9 T24OS ,WFCOS ,DTDOOS ,ETADOS ,WAB23OS ,OPDUOS ,DTDOCF ,ETADCF ,
AT7DS ,WFAOS ,UTAFOS ,ETAADS ,WAG6COS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,AT ,A8 ,A9 ,A28 ,A29
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,ABS AV ,A9SAV ,A28SAV ,A29SAV
COMMON / EACK/
1 T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2 WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,
3 T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4 T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5 WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6 FS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7 TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8 VA ,FRD ,VJD ,FGHD ,VJM ,FGHM ,FGPD ,FGPM ,
9 FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DATA ANORD/6HMMNOZZL/
WORD=ANORD
ABS AV=A8
A9SAV=A9
NOZM=0
IMNOZ=0
IF (NOZFLT.EQ.1.OR.NOZFLT.EQ.3) NOZM=1
IF (IGES.EQ.1.OR.IAFTBN.GT.0.OR.NOZM.EQ.1) IMNOZ=1
IF (IMCD.EQ.1) GO TO 1
CALL CONVRG(T7,H7,P7,S7,FAR7,WG7,F1,IMNOZ,A8, P7R,
1 T8,H8,P8,S8,TS8,PS8,V8,AM8,ICON)
GO TO (3,3,3,2),ICON
1 CALL CONDIV(T7,H7,P7,S7,FAR7,WG7,F1,IMNOZ,A8,A9,P7R,
1 T8,H8,P8,S8,T9,H9,P9,S9,TS8,TS9,PS8,PS9,V8,V9,AM8,AM9,ICON)
IMSHOC=ICON
GO TO (4,4,4,2),ICON
2 CALL ERROR
3 T9=T8
H9=H8
P9=P8
S9=S8
TS9=TS8
PS9=PS8
V9=V8
AM9=AM8
A9=A8
IMSHOC=ICON+3

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```
4      ERR(6)=(P7R-F7)/P7R
      IF(IMAGZ.EQ.1) WRITE(6,100) A8,AM8,A9,AM9
100    FORMAT(14HNOZZLE DESIGN,10X8H      A8=,E15.0,8H      AM8=,E15.8,
16H      A9=,E15.8,8H      AM9=,E15.8)
      RETURN
      END
```

```

SUBROUTINE PERF
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IANTP ,
2IGASHX ,IDURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,INSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOHAP ,NUMHAP ,HAPEDG ,TOLALL ,ERR (9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,OLMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIOS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,FRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFEDS ,OTCOOS ,ETABOS ,WA3CDS ,OPCOOS ,OTCOCF ,ETABCF ,
6TFHPDS ,CNHPCS ,ETHPCS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7YFIPDS ,CNIPCS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIFCF ,T21DS ,
8TFLPDS ,CNLPOS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T240S ,WFODS ,OTDUOS ,ETADDS ,WA23OS ,OPDUOS ,OTDUCF ,ETADCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,OPAFDS ,DTAFCF ,ETAACF ,
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A25 ,A29 ,
CPS55 ,AH55 ,CVNOZ ,CVHNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRCNT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,FRF ,ETAF ,WAF ,WAF ,BLDUI ,BLOUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCCN ,WG4 ,
9CNHP ,ETATHF ,DHTCHP ,DHTC ,BLHP ,BLHFI ,BLHFC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIF ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPC3LOUC ,PCBLCBI ,PCBLCBI ,PCBLHFI ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLIPI ,
GPCBLIPC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XELF ,X3LOU ,XBLOUI ,XBLOUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6HAD ,WFC ,WG24 ,FAR24 ,ETAO ,OPDUC ,BYPASS ,DUMS7 ,
7TS24 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON/ EACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXWER ,XWG55 ,XFAR55 ,XWFO ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,ANG ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGHM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DATA AWCRO/6H FERF/
WORD = AWORD

```

```

G=32.174049
CAPSF=2116.2170
WFT=WFB+WFD+WFA
WAT=WAF-BLQB
WGT=WAT+WFE
FART=WFT/WAT
VA=AM*CS
FRD=VA*WAF/G
VJM=CVMNOZ*V9
FGMH=VJM*WG7/G
FGPM=CAPSF*(PS9-P1)*A9
IF (IGASHX.GT.0) GO TO 1
VJD=CVDNOZ*V29
FGMD=VJD*WG24/G
FGPD=CAPSF*(FS29-P1)*A29
1 FGM=FGMH+FGMD
FGP=FGPM+FGPD
FG=FGP+FGP
FN=FG-FRD
SFC=3600.*WFT/FN
FG=DELFG*FG
FN=DELFN*FN
SFC=DELSFC*SFC
CALL CUTPUT
CALL ERROR
RETURN
END

```

```

SUBROUTINE CCNOUT(ICON)
  DIMENSION WORDY(345), IOUT(150), AOUT(6), WOUT(6), PARA1(96),
  1 PARA2(121), PARAM3(56), PARAM4(72)
  COMMON / ALL/
  1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IANTP ,
  2IGASHX, I'DBURN, I'AFTBN, IDCO ,IMCO ,IDSHOC, IMSHOC, NOZFLT,
  3ITRYS, LOOPER, NUNMAP, NUMMAP, MAPEOG, TOLALL, ERR(9)
  COMMON/DESIGN/
  1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
  2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
  3ZIOS ,PCNICS ,PRIOS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
  4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
  5T4DS ,WFBDS ,OTGDS ,ETABDS ,WA3CDS ,OPCOOS ,OTCCOF ,ETABCF ,
  6TFHPOS ,CNHPOS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
  7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
  8TFLPOS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
  9T24DS ,WFCDS ,OTDUDS ,ETAUDS ,WA23DS ,OPDUDS ,DYDUCF ,ETAOCF ,
  AT7DS ,WFADS ,OTAFUS ,ETAADS ,WG6CDS ,OPAFDS ,ETAFCF ,ETAACF ,
  BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
  CPS55 ,AM55 ,CVNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
  COMMON/ FRONT/
  1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
  2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
  3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
  4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
  5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
  6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
  7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
  8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCOM ,WG4 ,
  9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
  ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
  BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
  CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
  DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
  EWF8 ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLC ,
  FCBLC ,PCBLCBI, PCBLOC, PCBLPI, PCBHFC, PCBIFI, PCBIPC, PCBLLF ,
  GCBLLPC
  COMMON/ SIDE/
  1XP1 ,XWAF ,XWAI ,XWAC ,XELF ,XBLOU ,XBLOUI ,XBLOUC ,
  2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
  3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
  4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
  5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
  6W24 ,WFC ,WG24 ,FAR24 ,ETAD ,OPDUC ,BYPASS ,DUMS7 ,
  7T28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
  COMMON / BACK/
  XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
  XXWF8 ,XHG55 ,XFAR55 ,XWFO ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
  3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
  4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
  5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
  6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
  7T28 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
  8VA ,FRD ,VJD ,FGMD ,VJM ,FGMH ,FGPD ,FGPH ,
  9FAM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC

```

EQUIVALENCE (PARAM1(1),PCNFGU), (PARAM2(1),T1), (PARAM3(1),XP1),
1 (PARAM4(1),XT55)

DATA (WORDY(I),I=1,96)/

| | | | | | | |
|------------|-----------|-----------|-----------|-----------|------------|---|
| 17HPCNFGU | ,7HPCNIGU | ,7HPCNCGU | ,7HT4GU | ,7HDUM01 | ,7HDELFG | , |
| 27HDELFN | ,7HDELSFC | ,7HZFDS | ,7HPCNFOS | ,7HPRFOS | ,7HETAFOFS | , |
| 37HMAFOS | ,7HPRFCF | ,7HETAFCF | ,7HMAFCF | ,7HZIOS | ,7HPCNIOS | , |
| 47HPRIOS | ,7HETAIOS | ,7HMAIOS | ,7HPRICF | ,7HETAICF | ,7HMAICF | , |
| 57HZCOS | ,7HPCNCGS | ,7HPRCOS | ,7HETACOS | ,7HMACS | ,7HPRCCF | , |
| 67HETACCF | ,7HMACCF | ,7HT4DS | ,7HMFBS | ,7HDTCCDS | ,7HETABDS | , |
| 77HMA3COS | ,7HOPCOS | ,7HDTCCCF | ,7HETABCF | ,7HTFHPDS | ,7HCNHPDS | , |
| 87HETHFOS | ,7HTFHPCF | ,7HCNHPCF | ,7HETHPCF | ,7HDHHPCF | ,7HT2DS | , |
| 97HTFIPDS | ,7HCNIPDS | ,7HETIPDS | ,7HTFIPCF | ,7HCNIPCF | ,7HETIPCF | , |
| 17HDHIPCFC | ,7HT21DS | ,7HTFLPDS | ,7HCNLPDS | ,7HETLPCF | ,7HTFLPCF | , |
| 27HCNLPFC | ,7HETLPCF | ,7HDHLPFC | ,7HT22DS | ,7HT24DS | ,7HWFOS | , |
| 37HDTODS | ,7HETAADS | ,7HMA23DS | ,7HDPDUS | ,7HDTDUCF | ,7HETADCF | , |
| 47HT7OS | ,7HWFADS | ,7HDTAFOS | ,7HETAADS | ,7HWG6COS | ,7HOPAFOS | , |
| 57HDTAFCF | ,7HETAACF | ,7HA55 | ,7HA25 | ,7HA6 | ,7HA7 | , |
| 67HA8 | ,7HA9 | ,7HA28 | ,7HA29 | ,7HPS55 | ,7HAM55 | , |
| 77HCVDNOZ | ,7HCVHNOZ | ,7HA8SAV | ,7HA9SAV | ,7HA28SAV | ,7HA29SAV | / |

DATA (WORDY(I),I=97,150)/

| | | | | | | |
|---------|---------|---------|----------|----------|----------|---|
| 17HT1 | ,7HP1 | ,7HH1 | ,7HS1 | ,7HT2 | ,7HP2 | , |
| 27HH2 | ,7HS2 | ,7HT21 | ,7HP21 | ,7HH21 | ,7HS21 | , |
| 37HT22 | ,7HP22 | ,7HH22 | ,7HS22 | ,7HT3 | ,7HP3 | , |
| 47HH3 | ,7HS3 | ,7HT4 | ,7HP4 | ,7HH4 | ,7HS4 | , |
| 57HT45 | ,7HP45 | ,7HH45 | ,7HS45 | ,7HT5 | ,7HP5 | , |
| 67HH5 | ,7HS5 | ,7HT55 | ,7HP55 | ,7HH55 | ,7HS55 | , |
| 77HBLF | ,7HBLI | ,7HBLC | ,7HBLDU | ,7HCNF | ,7HPRF | , |
| 87HETAF | ,7HMAFC | ,7HMAF | ,7HBLDUI | ,7HBLDUC | ,7HBL0B | , |
| 97HCNI | ,7HPRI | ,7HETAI | ,7HMAIC | ,7HMAI | ,7HBL0BI | / |

DATA (WORDY(I),I=151,217)/

| | | | | | | |
|------------|------------|------------|------------|------------|------------|---|
| 17HBL0BC | ,7HMA3 | ,7HCNC | ,7HPRC | ,7HETAC | ,7HMACC | , |
| 27HWAC | ,7HETAB | ,7HOPCOM | ,7HWG4 | ,7HCNHP | ,7HETATHP | , |
| 37HDHTCMP | ,7HCHTC | ,7HBLHP | ,7HBLHPI | ,7HBLHPC | ,7HFAR4 | , |
| 47HCNIP | ,7HETATIF | ,7HDHTCIP | ,7HDHTI | ,7HBLIF | ,7HBLIPI | , |
| 57HBLIPC | ,7HCUMF | ,7HCNLP | ,7HETATLP | ,7HDHTCLP | ,7HDHTF | , |
| 67HBLLP | ,7HBLIPI | ,7HBLIPC | ,7HCS | ,7HWG45 | ,7HFAR45 | , |
| 77HWG5 | ,7HFAR5 | ,7HWG55 | ,7HFAR55 | ,7HHPEXT | ,7HAM | , |
| 87HALTP | ,7HETAR | ,7HZF | ,7HPCNF | ,7HZI | ,7HPCNI | , |
| 97HZC | ,7HPCNC | ,7HWF8 | ,7HTFFHP | ,7HTFFIP | ,7HTFFLP | , |
| 17HPCBLF | ,7HPCBLI | ,7HPCBLC | ,7HPCBLDUI | ,7HPCBLDUC | ,7HPCBL0BI | , |
| 27HPCBL0BC | ,7HPCBLHPI | ,7HPCBLHPC | ,7HPCBLIPI | ,7HPCBLIPC | ,7HPCBLLP | , |
| 37HPCBLLP | | | | | | / |

DATA (WORDY(I),I=218,273)/

| | | | | | | |
|-----------|-----------|----------|----------|-----------|----------|---|
| 17HXP1 | ,7HXWAF | ,7HXHA1 | ,7HXWAC | ,7HXBLF | ,7HXBLDU | , |
| 27HXBLDUI | ,7HXBLDUC | ,7HXH22 | ,7HXH3 | ,7HXT21 | ,7HXP21 | , |
| 37HXH21 | ,7HXS21 | ,7HDUMS1 | ,7HDUMS2 | ,7HT23 | ,7HP23 | , |
| 47HH23 | ,7HS23 | ,7HT24 | ,7HP24 | ,7HH24 | ,7HS24 | , |
| 57HT25 | ,7HP25 | ,7HH25 | ,7HS25 | ,7HT28 | ,7HP28 | , |
| 67HH28 | ,7HS28 | ,7HT29 | ,7HP29 | ,7HH29 | ,7HS29 | , |
| 77HDUMS3 | ,7HDUMS4 | ,7HDUMS5 | ,7HDUMS6 | ,7HWAU | ,7HWFU | , |
| 87HWG24 | ,7HFAR24 | ,7HETAD | ,7HOPDUC | ,7HBYPASS | ,7HDUMS7 | , |
| 97HTS28 | ,7HPS28 | ,7HV28 | ,7HAM28 | ,7HTS29 | ,7HPS29 | , |
| 17HV29 | ,7HAM29 | / | | | | |

DATA (WORDY(I),I=274,345)/

| | | | | | | |
|---------|---------|---------|---------|---------|---------|---|
| 17HXT55 | ,7HXP55 | ,7HXH55 | ,7HXS55 | ,7HXT25 | ,7HXP25 | , |
|---------|---------|---------|---------|---------|---------|---|


```

27HXH25 , 7HXS25 , 7HXWFB , 7HXWG55 , 7HXFAR55 , 7HXWFD ,
37HXHG24 , 7HXFAR24 , 7HXXP1 , 7HOLMB , 7HT6 , 7HP6 ,
47H46 , 7HS6 , 7HT7 , 7HP7 , 7HH7 , 7HS7 ,
57HT8 , 7HF8 , 7HH8 , 7HS8 , 7HT9 , 7HP9 ,
67H49 , 7HS9 , 7HWG6 , 7HWFA , 7HWG7 , 7HFAR7 ,
77HETA4 , 7HCPAFT , 7HV55 , 7HV25 , 7HPS6 , 7HV6 ,
87HAM6 , 7HTS7 , 7HPS7 , 7HV7 , 7HAM7 , 7HAM25 ,
97HTS8 , 7HFS8 , 7HV8 , 7HAM8 , 7HTS9 , 7HPS9 ,
17HV9 , 7HAM9 , 7HVA , 7HFRO , 7HVJD , 7HFGMD ,
27HJH , 7HFGMM , 7HFGPD , 7HFGPM , 7HFGM , 7HFGP ,
37HWFT , 7HWGT , 7HFART , 7HFG , 7HFN , 7HSFC /

```

```

DATA THEEND, BLANK, LIMIT, /7HTHEEND , 7H , , 345/

```

```

GO TO (1,12), ICCN

```

```

C *** INPUT SECTION

```

```

1 GO 4 N=1,150

```

```

NUM=N

```

```

READ(5,100)AIN,CHANGE

```

```

IF(AIN.EQ.THEEND) GO TO 5

```

```

DO 2 J=1,LIMIT

```

```

JJ=J

```

```

IF(AIN.EQ.WORDY(J)) GO TO 3

```

```

2 CONTINUE

```

```

WRITE(6,101)AIN

```

```

GO TO 4

```

```

3 IOUT(NUM)=JJ

```

```

IF(CHANGE.NE.BLANK) WORDY(JJ)=CHANGE

```

```

4 CONTINUE

```

```

WRITE(6,102)

```

```

5 NUM=NUM-1

```

```

RETURN

```

```

C *** OUTPUT SECTION

```

```

12 IF(NUM.EQ.1) GO TO 16

```

```

N=NUM

```

```

J=6

```

```

DO 15 I=1,NUM,6

```

```

IF(N.GT.6) GO TO 13

```

```

L=N

```

```

13 N=N-6

```

```

DO 14 K=1,J

```

```

L=L+K-1

```

```

M=IOUT(L)

```

```

WOUT(K)=WORDY(M)

```

```

IF(M.GT.96) GO TO 20

```

```

WOUT(K)=PARAM1(M)

```

```

GO TO 14

```

```

20 IF(M.GT.217) GO TO 21

```

```

MN=M-96

```

```

WOUT(K)=PARAM2(MN)

```

```

GO TO 14

```

```

21 IF(M.GT.273) GO TO 22

```

```

MN=M-217

```

```

WOUT(K)=PARAM3(MN)

```

```

GO TO 14

```

```

22 MN=M-273

```

```

WOUT(K)=PARAM4(MN)

```

```
14 CONTINUE
   WRITE(6,103) (WOUT(K),K=1,J)
   WRITE(6,104) (AOUT(K),K=1,J)
   IF(N.LE.0) GO TO 16
15 CONTINUE
16 RETURN
100 FORMAT(A7,7X,A7)
101 FORMAT('0H0THE WCRD ,A7,26H NOT FCUND IN COMMON ARRAY)
102 FORMAT(22H0ERROR IN CONOUT INPUT)
103 FORMAT(1H ,25XA7,5(8XA7))
104 FORMAT(1H ,20X6E15.6)
END
```

SUBROUTINE ERROR

DIMENSION TRASH1(96), TRASH2(121), TRASH3(56), TRASH4(72)

COMMON / ALL/

1 WORD , IDES , JDES , KDES , MODE , INIT , IDUMP , IAMP ,
 2 IGASHX, IOBURN, IAFTBN, IDCD , IMCD , IOSHOC, IMSHOC, NOZFLT,
 3 ITSYS, LOOPER, NOHAP, NUMMAP, MAPEOG, TOLALL, ERR(9)

COMMON/DESIGN/

1 PCNFGU , PCNIGU , PCNCGU , T4GU , DLMD1 , DELFG , DELFN , DELSFC ,
 2 PRFDS , PCNFDS , PRFDS , ETAFDS , WAFDS , PRFCF , ETAFCF , WAFCF ,
 3 PRIOS , PCNIOS , PRIOS , ETAIOS , WAIOS , PRICF , ETAICF , WAICF ,
 4 PCNCS , PCNCS , PCNCS , ETACDS , WACDS , PRCCF , ETACCF , WACCF ,
 5 WFBS , WFBS , DTGDS , ETABDS , WA3CDS , DPCDS , DTGCCF , ETABCF ,
 6 CNHPDS , CNHPDS , ETHPDS , TFHPCF , CNHPCF , ETHPCF , DHHPCF , T2DS ,
 7 CNIPDS , CNIPDS , ETIPDS , TFIPCF , CNIPCF , ETIPCF , DHIPCF , T2IDS ,
 8 CNLPS , CNLPS , ETLPDS , TFLPCF , CNLPCF , ETLPDF , DHLPCF , T22DS ,
 9 WFADS , WFADS , DTADDS , ETADDS , WA23DS , DPODS , DTDFCF , ETADCF ,
 A17DS , WFADS , DTAFDS , ETAADS , WG6CDS , DPAFDS , DTAFCF , ETAACF ,
 B45 , A25 , A6 , A7 , A8 , A9 , A28 , A29 ,
 C55 , AM55 , CVDNOZ , CVMNOZ , A8SAV , A9SAV , A28SAV , A29SAV

COMMON/ FRONT/

1 T1 , P1 , H1 , S1 , T2 , P2 , H2 , S2 ,
 2 T21 , P21 , H21 , S21 , T22 , P22 , H22 , S22 ,
 3 T3 , P3 , H3 , S3 , T4 , P4 , H4 , S4 ,
 4 T45 , P45 , H45 , S45 , T5 , P5 , H5 , S5 ,
 5 T55 , P55 , H55 , S55 , BLF , BLI , BLC , BLOU ,
 6 CNF , PRF , ETAF , WAF , WAF , BLDUI , BLDUC , BLOB ,
 7 CNF , PRI , ETAI , WAIC , WAI , BLOBI , BLOB , WA3 ,
 8 CNF , PRC , ETAC , WACC , WAC , ETAB , DPCOM , WG4 ,
 9 CNF , ETATHP , DHTCHP , DHTC , BLHP , BLHPI , BLHPC , FAR4 ,
 ACNIP , ETATIP , DHTCIP , DHTI , BLIP , BLIPI , BLIPC , DUMF ,
 BCNLP , ETATLP , DHTCLP , DHTF , BLLP , BLLPI , BLLPC , CS ,
 CH45 , FAR45 , WG5 , FAR5 , WG55 , FAR55 , HPEXT , AM ,
 DALTP , ETAR , ZF , PCNF , ZI , PCNI , ZC , PCNC ,
 EMFB , TFFHF , TFFIP , TFFLP , PCBLF , PCBLI , PCBLC , PCBLDT ,
 FBLOUC , PCLOBI , PCLOBC , PCBLHI , PCBLHPC , PCBLIFI , PCBLIPC , PCBLPI ,
 GPBLFC

COMMON/ SIDE/

1 XP1 , XWAF , XWAI , XWAC , XBLF , XBLOU , XBLOUI , XBLOUC ,
 2 XH21 , XH3 , XT21 , XP21 , XT21 , XS21 , DUMS1 , DUMS2 ,
 3 T23 , P23 , H23 , S23 , T24 , P24 , H24 , S24 ,
 4 T25 , P25 , H25 , S25 , T28 , P28 , H28 , S28 ,
 5 T29 , P29 , H29 , S29 , DUMS3 , DUMS4 , DUMS5 , DUMS6 ,
 6 W21 , WFC , WG24 , FAR24 , ETAD , DPOUC , BYPASS , DUMS7 ,
 7 T28 , PS28 , V28 , AM28 , TS29 , PS29 , V29 , AM29

COMMON/ EACK/

XX55 , XP55 , XH55 , XS55 , XT25 , XP25 , XH25 , XS25 ,
 XX55 , XHG55 , XFAR55 , XWFD , XHG24 , XFAR24 , XXP1 , DUMB ,
 3 T6 , P6 , H6 , S6 , T7 , P7 , H7 , S7 ,
 4 T8 , P8 , H8 , S8 , T9 , P9 , H9 , S9 ,
 5 WG6 , WFA , WG7 , FAR7 , ETAA , DFAFT , V55 , V25 ,
 6 PS6 , V6 , AM6 , TS7 , PS7 , V7 , AM7 , AM25 ,
 7 T8 , PS8 , V8 , AM8 , TS9 , PS9 , V9 , AM9 ,
 8 VA , FRD , VJD , FGMD , VJM , FGMM , FGPD , FGFM ,
 9 FCM , FGP , HFT , HGT , FART , FG , FN , SFC

EQUIVALENCE (TRASH1(1), PCNFGU), (TRASH2(1), T1), (TRASH3(1), XP1)

```

EQUIVALENCE (TRASH4(1),XT55)
DATA AWORD/6HCOMMON/
WRITE(6,100)WORD
WORD=AWORD
WRITE(6,102)WORD,ZF,PCNF,ZI,PCNI,ZC,PCNC,T4,MODE
WRITE(6,103)
WRITE(6,104)(TRASH1(I),I=1,96)
WRITE(6,105)
WRITE(6,104)(TRASH2(I),I=1,121)
WRITE(6,105)
WRITE(6,104)(TRASH3(I),I=1,56)
WRITE(6,103)
WRITE(6,104)(TRASH4(I),I=1,72)
WRITE(6,103)
WRITE(6,106)LOOPER
IF(IDUMP.EQ.0)GO TO 2
WRITE(6,105)
CALL SYG(2)
2 CALL ENGBAL
RETURN
100 FORMAT(28H0AN ERROR HAS BEEN FOUND IN ,A6)
102 FORMAT(1H0,A6,9X,7E15.6,I4)
103 FORMAT(2H0 )
104 FORMAT(1H0,8E15.6)
105 FORMAT(1H1)
106 FORMAT(25H0FAILED TO CONVERGE AFTER,I4,6H LOOPS)
END

```

```

SUBROUTINE SYG(ICON)
DIMENSION WORD(132)
DATA ONEDOL/6H$ /
GO TO (1,2),ICON
1  END FILE 8
   REWIND 8
   RETURN

C      TERMINATE THE FILE

2  WRITE(8,500)
500 FORMAT(12H$$$$$$$$$$$$)
   END FILE 8
   REWIND 8

C      READ RECORD

5  READ(8,501) (WORD(I),I=1,132)
501 FORMAT(132A1)

C      CHECK FOR 12 LEADING DOLLAR SIGNS

DO 10 I=1,12
IF (WORD(I)-ONEDOL)11,10,11
10 CONTINUE
   RETURN

C      CHECK FOR 6 TRAILING DOLLAR SIGNS
11 DO 15 I=1,132

IF (WORD(I)-ONEDOL)15,12,15
12 K=I+5
DO 13 J=I,K
IF (WORD(J)-ONEDOL)15,13,15
13 CONTINUE
   GO TO 20
15 CONTINUE
WRITE(6,502)
502 FORMAT(1H0,12HERROR IN SYG)
   RETURN

      PRINT LINE

0  I=I-1
   WRITE(6,505) (WORD(M),M=1,I)
505 FORMAT(132A1)
   GO TO 5
   END

```

```
SUBROUTINE THCOMP(PR,ETA,T,H,S,P,TO,H0,SO,PO)
PO=P*FR
TP=T*PR*.28572
DO 1 I=1.25
CALL THERMC(FO,HP,TP,SP,X1,0,X2,0)
DELS=SP-S
IF (ABS(DELS).LE.0.00005*S) GO TO 2
1 TP=TP/EXP(4.*DELS)
CALL ERROR
2 HO=H+((HF-H)/ETA)
CALL THERMC(FO,HO,TO,SO,X1,0,X2,1)
RETURN
END
```

SUBROUTINE THTURB(DH,ETA,FAR,H,S,F,TC,HC,SO,PC)

HO=H-CH

HOP=H-DH/ETA

PT=P/2.

DO 1 I=1,25

CALL THERMC(PT,HOP,TT,ST,AMWT,1,FAR,1)

DELS=ST-S

IF (ABS(DELS).LE.0.00005*S) GO TO 2

1 PT=P*EXP(DELS*AMWT/1.986375+ALOG(FT/P))

CALL ERROR

2 PO=PT

CALL THERMC(PO,HC,TO,SO,X1,1,FAR,1)

RETURN

END

```

SUBROUTINE PRCCOM(FARX,TCX,CSEX,REX,CPEX,REX,PHI,HEX)
IF (FARX.LE.0.0) GO TO 1
FARX=0.067623
WRITE(8,101)
1 IF (TEX.GE.300.) GO TO 2
  TLX=300.
  WRITE(8,102)
2 IF (TEX.LE.4000.) GO TO 3
  TLX=4000.
  WRITE(8,103)
3 IF (FARX.GE.0.0) GO TO 4
  FARX=0.0
  WRITE(8,104)
C AIR PATH
4 CPA = (((((1.0115540E-25*TEX-1.4526770E-21)*TEX
1+7.621577E-18)*TEX-1.5122259E-14)*TEX-6.7178376E-12)
2*TEX+6.5519486E-08)*TEX-5.1536879E-05)*TEX+2.5020051E-01
  HEA = ((((((1.2644425E-26*TEX-2.752522E-22)*TEX
1+1.2702630E-18)*TEX-3.0256518E-15)*TEX-1.6794594E-12)*TEX
2+2.1839826E-08)*TEX-2.573440E-05)*TEX+2.5020051E-01)*TEX
3-1.755883E-01
  SEA = +2.010091E-01*ALOG(TEX) + ((((((1.4450767E-26*TEX
1-2.4211288E-22)*TEX+1.5243153E-18)*TEX-3.7820648E-15)*TEX
2-2.2392790E-12)*TEX+3.2759743E-08)*TEX-5.1576679E-05)*TEX
3+4.5432300E-02
  IF (FARX.LE.0.0) GO TO 5
C FUEL/AIR PATH
CPF = ((((((7.2678710E-25*TEX-1.3335668E-20)*TEX
1+1.1212913E-16)*TEX-4.2051104E-13)*TEX+3.9686793E-10)*TEX
2-1.3771901E-06)*TEX+1.2258630E-03)*TEX+7.3816638E-02
  HEF = ((((((9.0848308E-26*TEX-1.9050949E-21)*TEX
1+1.7021525E-17)*TEX-8.4102208E-14)*TEX+2.4921698E-10)*TEX
2-4.5906332E-07)*TEX+6.1293150E-04)*TEX+7.3816638E-02)
3*TEX+3.0501530E+01
  SEF = +7.3816638E-02*ALOG(TEX) + ((((((1.0382670E-25*TEX
1-2.222611E-21)*TEX+2.0425326E-17)*TEX-1.0512776E-13)*TEX
2+3.3228928E-10)*TEX-6.885950E-07)*TEX+1.2258630E-03)*TEX
3+6.483399E+01
5 CPEX = (CPA+FARX)/(1.+FARX)
  HEX = (HEA+FARX*HEF)/(1.+FARX)
  PHI = (SEA+FARX*SEF)/(1.+FARX)
  AMW = 28.97-.946106*FARX
  RWX = 1.98637E-01
  REX = CPE / (CPE + REX)
  CSEX = SQRT(AMW*REX*TLX*25031.37)
  RETURN
101 FORMAT(1H,63HINPUT FUEL-AIR RATIO ABOVE LIMITS, IT HAS BEEN RESET
2TO 0.067623,6H$$$$$$)
102 FORMAT(1H,63HPRCCOM INPUT TEMPERATURE BELOW 300.,6H$$$$$$)
103 FORMAT(1H,63HPRCCOM INPUT TEMPERATURE ABOVE 4000.,6H$$$$$$)
104 FORMAT(1H,38HPRCCOM INPUT FUEL-AIR RATIO BELOW ZERO,6H$$$$$$)
  END

```



```

SUBROUTINE SEARCH(P,A,B,C,D,AX,NA,BX,CX,DX,NO,NAH,NGH,NCODE)
DIMENSION AX(NAH),BX(NAH,NOH),CX(NAH,NOH),DX(NAH,NOH),NO(NAH),Q(9)
C *** NEEDS SUBROUTINE AFQUIR
C *** AX AND BX MUST BE STORED LO TO HI
C *** P= INPUT PROPORTION BETWEEN 0.0 AND 1.0
C *** IF NOT INPUT, P MUST EQUAL -1.
C *** NCODE=00 OK
C NCODE=01 A LO
C NCODE=02 A HI
C NCODE=07 ERROR
C NCODE=10 B LO
C NCODE=20 B HI
C NCODE=0
C=0.
D=0.
C *** FIND A
DO 1 I=1,NA
IH=I
IF (A.LT.AX(I)) GO TO 2
1 CONTINUE
IF (A.GT.AX(IH)) NCODE=2
A=AX(IH)
GO TO 3
2 IF (IH.GT.1) GO TO 3
NCODE=1
IH=2
A=AX(1)
3 IL=IH-1
LIH=NO(IH)
LIL=NO(IL)
C *** FIND B
FRM=(A-AX(IL))/(AX(IH)-AX(IL))
PP=0
IF (P.GE.0.) GO TO 6
BL=AX(IL,1)+FRM*(BX(IH,1)-BX(IL,1))
BH=AX(IL,LIL)+FRM*(BX(IH,LIH)-BX(IL,LIL))
IF (P.GE.BL) GO TO 4
NCODE=NCODE+10
B=BL
GO TO 5
4 IF (P.LE.BH) GO TO 5
NCODE=NCODE+20
B=B-
5 PP=0.5
C(2)=0.
Q(2)=0.
6 BH=PP*(BX(IH,LIH)-BX(IH,1))+BX(IH,1)
BL=PP*(BX(IL,LIL)-BX(IL,1))+BX(IL,1)
DO 7 J=2,LIMH
JH=J
IF (BH.LT.BX(IH,J)) GO TO 8
7 CONTINUE
8 JL=JH-1
GO 9 K=2,LIML
KH=K

```

```

IF (BL.LT.BX(IL,K)) GO TO 10
9 CONTINUE
10 KL=KH-1

PR=(BX(IH,JL)-BH)/(BX(IH,JH)-BX(IH,JL))
CH=BX(IH,JL)-PR*(CX(IH,JH)-CX(IH,JL))
DH=DX(IH,JL)-PR*(DX(IH,JH)-DX(IH,JL))

PR=(BX(IL,KH)-BL)/(BX(IL,KH)-BX(IL,KL))
CL=BX(IL,KH)-PR*(CX(IL,KH)-CX(IL,KL))
DL=DX(IL,KH)-PR*(DX(IL,KH)-DX(IL,KL))

BT=BL+PR*(CH-CL)
CT=CL+PR*(DH-CL)
DT=DL+PR*(DH-DL)

IF (P.GT.1) GO TO 13
DIR=SORT(B/BT)
ERR=(C-B)/B
CALL AFQUIR(Q(1),PP,ERR,0.,25.,0.001,DIR,PT,ICON)
GO TO (11,12,12),ICON
PP=PT
IF (PP.LT.0.) PP=0.
IF (PP.GT.1) PP=1.
GO TO 6
12 MCODE=7
13 B=BT
C=CT
D=DT
RETURN
END

```

```

SUBROUTINE CCNVRG(TI,HI,PI,SI,FAR,WG,PA,IDES,AO,PR,
1 TO,PO,SC,TSO,PSO,VO,AMO,ICON)
C   ICON=1      SUBSONIC, COMPARE PI WITH PR
C   ICON=2      SONIC, COMPARE PI WITH PR
C   ICON=4      ERROR
AJ=778.26
CAPSF=2116.217
G=32.174049
CALL PROCOM(FAR,TI,XX1,XX2,XX3,XX4,PHI,XX6)

```

C *** SONIC CALCULATIONS

```

J=0
TSS=0.833*TI
1  J=J+1
   CALL PROCOM(FAR,TSS,CSS,AKS,CP,REXS,PHISS,HSS)
   HSCAL=HI-CSS**2/(2.*G*AJ)
   DELHS=HSCAL-HSS
   IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2
2  TSS=TSS+DELHS/CF
   IF (J-15) 1,1,3
3  ICON=4
   RETURN
4  IF (IDES) 12,12,5

```

C *** ISENTROPIC EXPANSION CALCULATIONS

```

5  J=0
   TSI=TI*(PA/PI)**0.286
6  J=J+1
   CALL THERMO(PA,HSI,TSI,SSI,XX1,1,FAR,0)
   IF (ABS(SSI-SI)-0.0001*SI) 8,8,7
7  TSI=TSI/EXP(4.*(SSI-SI))
   IF (J-30) 6,6,3
8  VTS=SQRT(2.*G*AJ*(HI-HSI))
   IF (HSI-CSS) 9,11,11

```

C *** SUBSONIC DESIGN, CALCULATE AO

```

9  VO=VIS
   TSO=TSI
   FSC=PA
   CALL PROCOM(FAR,TSO,CSO,XX2,XX3,REX,PHISO,HSO)
   RHO=(CAPSF*FSC/(AJ*REX*TSO))
   AO=WG/(RHO*VO)
   AMO=VO/CSO
   PP=PT
   ICON=1
10 TO=TI
   HO=HI
   PO=PI
   SO=SI
   RETURN

```

C *** SONIC DESIGN, CALCULATE AO

```

11  VO=CSS
    TSO=TSS
    PSO=PI*(TSO/TSI)**(AKS/(AKS-1.))
    RHO=CAPSF*FSC/(AJ*REXS*TSO)
    AO=WG/(RHO*VC)
    AMO=1.0
    PR=PI
    ICON=2
    GO TO 10

```

C *** NON-DESIGN, CALCULATE CRITICAL CONDITIONS

```

12  VO=CSS
    TSO=TSS
    PSO=PA
    RHO=CAPSF*PSO/(AJ*REXS*TSO)
    AOCRIT=WG/(RHO*VO)
    AMO=1.0
    PR=PSO*(TI/TSO)**(AKS/(AKS-1.))
    IF(AO-AOCRIT)13,13,14

```

C *** NON-DESIGN, CRITICAL AND SUPERCRITICAL CONDITIONS

```

13  FSO=PSO*AOCRIT/AO
    PR=PR*AOCRIT/AO
    ICON=2
    GO TO 10

```

C *** NON-DESIGN, SUBSONIC CALCULATIONS

```

14  PSO=PA
    J=0
    TSO=0.833*TSO
15  J=J+1
    CALL PROCOM(FAR,TSO,CSO,AKO,CP,REX,PHISO,HSO)
    RHO=CAPSF*FSC/(AJ*REX*TSO)
    VO=WG/(RHO*AO)
    HSCAL=HI-VC**2/(2.*G*AJ)
    DELHS=HSCAL-HSO
    IF(ABS(DELHS)-0.0005*HSCAL)17,17,16
16  TSO=TSO+DELHS/CP
    IF(J-15)15,15,3
17  AMO=VO/CSO
    PR=PSO*(TI/TSO)**(AKO/(AKO-1.))
    ICON=1
    GO TO 10
    END

```

```

SUBROUTINE CCNDIV(TI,PI,PI,SI,FAR,WG,PA,IDES,AT,AO,FIR,
1 TI,HT,PT,ST,TC,HO,PO,SO,TST,TSO,PST,PSO,VT,VO,AMT,AMO,ICON)
C   ICON=1  SUBSONIC, COMPARE PIR WITH PI
C   ICON=2  SONIC, SHOCK INSIDE NOZZLE, COMPARE PIR WITH PI
C   ICON=3  SONIC, SHOCK OUTSIDE NOZZLE, COMPARE PIR WITH PI
C   ICON=4  ERROR
DIMENSION G(9)
      G(2)=0.
      G(3)=0.
      AJ=778.26
      CAPSF=2116.2170
      G=32.174049
      CALL PROCOM(FAR,TI,XX1,XX2,XX3,XX4,PHII,XX6)

C *** SONIC CALCULATIONS

      J=J
      TSS=0.833*TI
1     J=J+1
      CALL PROCOM(FAR,TSS,CSS,AK,CP,REXS,PHISS,HSS)
      HSCAL=HI-CSS**2/(2.*G*AJ)
      DELHS=HSCAL-TSS
      IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2
2     TSS=TSS+DELHS/CP
      IF (J-15) 1,1,3
3     ICON=4
      RETURN
4     IF (IDES) 11,11,5

C *** SONIC DESIGN, CALCULATE AT

5     VT=CSS
      TST=TSS
      PST=PI*(TST/TI)**(AK/(AK-1.))
      RHO=CAPSF*FST/(AJ*REXS*TST)
      AT=WG/(RHO*VT)
      AMT=1.0

C *** IDEAL EXPANSION DESIGN, CALCULATE AO

      PSC=PA
      J=0
      TSO=TI*(PSC/PI)**.286
6     J=J+1
      CALL PROCOM(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
      PHICAL=PHII-REX*ALOG(PI/PSO)
      DELPHI=PHICAL-PHISO
      IF (ABS(DELPHI)-0.0001*PHICAL) 8,8,7
7     TSO=TSO*EXP(4.*DELPHI)
      IF (J-15) 6,6,3
8     VO=SQRT(2.*G*AJ*(HI-HSO))
      AMO=VO/CSO
      AO=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*
1 (AK-1.)))
      FIR=PI

```

```

      ICON=3
9      TO=TI
      HO=HI
      PO=PI
      SO=SI
10     TT=TI
      HT=HI
      FT=PI
      ST=SI
      RETURN

C *** ASSUME SONIC THROAT AND ISENTROPIC EXPANSION TO AO

11     VT=...
      AMO=1.0
      P=1SS
      RHO=HG/(AT*VT)
      PST=RHO*AJ*REXS*TST/CAPSF
      PIR=PST*(TI/TST)**(AK/(AK-1.))
      IF (PST-PA) 12,24,24
12     TSO=0.95*TI
      MAM=0
13     CALL FROCOF(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
      AMO=SQRT(2.*((TI/TSO)-1.)/(AK-1.))
      AOAL=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/
1 (2.*(AK-1.)))
      EA=(AO-AOAL)/AC
      DIR=SQRT(AO/AOAL)
      CALL AFQUIR(Q(1),TSO,EA,0.,100.,0.0001,DIR,TSOT,JCON)
      GO TO (14,15,3),JCON
14     TSO=TSOT
      IF (TSC-TI) 140,13,141
140    TSC=2.*TI/(AK+1.)
      IF (TSO.GT.TSC) GO TO 142
141    TSO=0.98*TI
      GO TO 13
142    IF (Q(2).LT.30.0.OR.AMO.LT.0.95.OR.MAM.EQ.1) GO TO 13
      TSO=2.*TI/(2.+0.98*(AK-1.))
      MAM=1
      GO TO 13
15     FSO=PIR*(TSO/TI)**(AK/(AK-1.))
      IF (PSO-PA) 17,16,24
C *** CRITICAL FLOW, ISENTROPIC EXPANSION TO PA

16     VO=AMO*CSO
      ICON=1
      GO TO 9

C *** SUBSONIC FLOW

17     FSO=PA
      Q(2)=0.
      Q(3)=0.
      J=0
      TSO=0.833*TI

```

```

18  J=J+1
    CALL PROCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
    RHO=CAPSF*PSC/(AJ*REX*TSO)
    VO=HG/(RHO*AO)
    HSCAL=HI-VC**2/(2.*G*AJ)
    DELHS=HSCAL-HSO
    IF (ABS(DELHS)-0.0005*HSCAL) 20,20,19
19  TSO=TSO+DELHS/CF
    IF (J-15) 18,18,3
20  AMO=VO/CSO
    FIR=PSO*(TI/TSO)**(AK/(AK-1.))
    TST=TSO
21  CALL PROCOP(FAR,TST,CST,AK,CP,REX,PHIST,HST)
    PST=PIR*(TST/TI)**(AK/(AK-1.))
    RHO=PST*CAPSF/(AJ*REX*TST)
    VT=HG/(RHO*AT)
    HSCAL=HI-VT**2/(2.*G*AJ)
    EH=(HSCAL-HST)/HSCAL
    DIR=1.+(HSCAL-HST)/(CP*TST)
    CALL AFQUIR(Q(1),TST,EH,0.,20.,0.0005,DIR,TSTT,JCON)
    GO TO (22,23,3),JCON
22  TST=TSTT
    GO TO 21
23  AMT=VT/CST
    ICON=1
    GO TO 9

```

C *** SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO PA

```

24  PSO=PA
    J=0
    TSO=TI*(PSC/FIR)**.286
25  J=J+1
    CALL PROCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
    PHICAL=PHII-REX*ALOG(PIR/PSO)
    DELPHI=PHICAL-PHISO
    IF (ABS(DELPHI)-0.0001*PHICAL) 27,27,26
26  TSO=TSO*EXP(4.0*DELPHI)
    IF (J-15) 25,25,3
27  VO=SQRT(2.*G*AJ*(HI-HSO))
    AMO=VC/CSO
    AOI)=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/
1 (2.*(AK-1.)))
    ICON=3
    N=0
    IF (AO-AOI) 28,9,29

```

C *** SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO AO

```

28  N=1
29  TSO=0.833*TI
    J=0
30  J=J+1
    CALL PROCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
    AMO=SQRT(2.*(TI/TSO)-1.)/(AK-1.)

```

```

AOCAL=(AT/AMC)*(2.*(1+(AK-1.)*AMC**2/2.)/(AK+1.))**((AK+1.)/
1(2.*(AK-1.)))
CELA=AO-AOCAL
IF (ABS(CELA)-0.0001*AO) 32,32,31
31 TSO=TSO*SQRT(AOCAL/AO)
IF (J-50) 30,30,3
32 IF (N) 34,34,33

C *** UNDEREXPANDED, SHOCK OUTSIDE NOZZLE
33 PSQ=PIR*(TSO/TI)**(AK/(AK-1.))

VO=AMC*CSO
GO TO 9

C *** OVEREXPANDED, FIND SHOCK POSITION
34 PSX=PIR*(TSO/TI)**(AK/(AK-1.))
PSY=PSX*(2.*AK*AMC**2/(AK+1.)-(AK-1.)/(AK+1.))
IF (PA-PSY) 35,36,36

C *** OVEREXPANDED, SHOCK OUTSIDE NOZZLE
35 PSO=PSX
VO=AMC*CSO
GO TO 9

C *** OVEREXPANDED, SHOCK INSIDE NOZZLE
36 PSO=PA
J=0
TSO=0.833*TI
37 J=J+1
CALL PRUCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
RHO=CAPSF*PSC/(AJ*REX*TSO)
VO=HG/(RHO*AC)
HSCAL=HI-VO**2/(2.*G*AJ)
DELHS=HSCAL-HSO
IF (ABS(DELHS)-0.0005*HSCAL) 39,39,38
38 TSO=TSO+DELHS/CF
IF (J-15) 37,37,3
39 AMO=VC/CSO
TO=TI
HO=HI
PO=PSO*(TO/TSO)**(AK/(AK-1.))
SO=PHII-REX*ALOG(PO)
ICON=2
GO TO 10
END

```



```

SUBROUTINE FQUIR(X,AIND,DEPEND,ANS,AJ,TOL,DIR,ANEH,ICON)
DIMENSION X(9)
C X(1)=NAME OF ARRAY TO USE
C AIND=INDEPENDANT VARIABLE
C DEPEND= DEPENDANT VARIABLE
C ANS=ANSWER UPON WHICH TO CONVERGE
C AJ=MAX NUMBER OF TRYS
C TOL=PERCENT TOLERANCE FOR CONVERGENCE
C DIR=DIRECTION AND PERCENTAGE FOR FIRST GUESS
C ANEH=CALCULATED VALUE OF NEXT TRY AT INDEPENDANT VARIABLE
C ICON=CONTROL  =1 GO THRU LOOP AGAIN
C              =2 YOU HAVE REACHED THE ANSWER
C              =3 COUNTER HAS HIT LIMITS
C X(2)=COUNTER STORAGE
C X(3)=CHOOSES METHOD OF CONVERGENCE
C X(4)=THIRD DEPEND VAR
C X(5)=THIRD IND VAR
C X(6)=SECOND DEPEND VAR
C X(7)=SECOND INC VAR
C X(8)=FIRST DEPEND VAR
C X(9)=FIRST IND VAR
C X(3) MUST BE ZERO UPON FIRST ENTRY TO ROUTINE

```

```

      Y=0.
      IF (ANS)1,2,1
1     DEP=DEPEND-ANS
      TOLANS=TOL*ANS
      GO TO 3
2     DEP=DEPEND
      TOLANS=TOL
3     IF (ABS(DEP)-TOLANS)5,5,4
4     IF (X(2)-AJ)6,8,7
5     ANEH=AIND
      X(2)=0.
      ICON=2
      RETURN
6     ANEH=Y
      X(2)=X(2)+1.
      ICON=1
      RETURN
7     ANEH=Y
      X(2)=0.
      ICON=3
      RETURN
8     IF (X(3))9,9,12
C *** FIRST GUESS USING DIR
9     X(3)=1.
      X(8)=DEP
      X(9)=AIND
      IF (AIND)10,11,10
10    Y=DIR*AIND
      GO TO 6
11    Y=DIR
      GO TO 6
12    IF (X(3)-1.)13,13,16

```

```

C *** LINEAR GUESS
13  X(3)=2.
    X(6)=DEP
    X(7)=AIND
    IF (X(8)-X(6))14,9,14
14  IF (X(9)-X(7))15,9,15
15  A=(X(9)-X(7))/(X(8)-X(6))
    Y=X(9)-A*X(8)
    IF (ABS(10.*X(9))-ABS(Y))9,9,6
C *** QUADRATIC GUESS
16  X(4)=DEP
    X(5)=AIND
    IF (X(7)-X(5))18,17,18
17  IF (X(6)-X(4))13,9,13
18  IF (X(6)-X(4))19,13,19
19  IF (X(9)-X(5))23,20,23
20  IF (X(8)-X(4))21,22,21
21  X(9)=X(7)
    X(8)=X(6)
    GO TO 13
22  X(9)=X(7)
    X(8)=X(6)
    X(3)=1.
    IF (X(9))10,11,10
23  IF (X(8)-X(4))24,21,24
24  F=(X(6)-X(4))/(X(7)-X(5))
    A=(X(8)-X(4)-F*(X(9)-X(5)))/((X(9)-X(7))*(X(9)-X(F)))
    B=F-A*(X(5)+X(7))
    C=X(4)+X(5)*A*(X(7)-X(5))
    IF (A)242,240,242
240 IF (B)241,7,241
241 Y=-C/B
    GO TO 37
242 IF (B)247,243,247
243 IF (C)245,244,245
244 Y=0.
    GO TO 37
245 G=-C/A
    IF (G)7,7,246
246 Y=SQRT(G)
    YY=-SQRT(G)
    GO TO 270
247 IF (C)249,248,249
248 Y=-B/A
    YY=0.
    GO TO 270
249 D=4.*A*C/B**2
    IF (1.-D)13,25,26
25  Y=-B/(2.*A)
    GO TO 37
26  E=SQRT(1.-D)
27  Y=(-B/(2.*A))*(1.+E)
    YY=(-B/(2.*A))*(1.-E)
270 J=4
    DEPMIN=ABS(X(4))

```

```

DO 29 I=6, 8, 2
IF (DEPMIN-ABS(X(I))) 29, 29, 28
28 J=I
DEPMIN=ABS(X(I))
29 CONTINUE
K=J+1
IF ((X(K)-Y)*(X(K)-YY)) 32, 32, 30
30 IF (ABS(X(K)-Y)-ABS(X(K)-YY)) 37, 37, 31
31 Y=YY
GO TO 37
32 IF (J-6) 33, 34, 34
33 JJ=J+2
KK=K+2
GO TO 35
34 JJ=J-2
KK=K-2
35 SLOPE=(X(KK)-X(K))/(X(JJ)-X(J))
IF (SLOPE*X(J)*(X(K)-Y)) 36, 36, 37
36 Y=YY
37 X(9)=X(7)
X(8)=X(6)
X(7)=X(5)
X(6)=X(4)
GO TO 6
END

```

```

SUBROUTINE MATRIX(E,V,A)
DIMENSION E(9,9),V(9),A(9),PIV(10),T(9,10)
DO 1 I=1,9
T(I,10)=A(I)
DO 1 J=1,9
1 T(I,J)=E(I,J)
DO 7 I=1,9
TEMP=0.
DO 2 J=I,9
IF(TEMP.GT.ABS(T(J,I))) GO TO 2
TEMP=ABS(T(J,I))
IPIV=J
2 CONTINUE
IP1=I+1
DO 3 J=IP1,10
3 PIV(J)=T(IPIV,J)/T(IPIV,I)
IFROM=9
ITO=9
4 IF(IFROM.EC.IPIV) GO TO 6
RM=-T(IFROM,I)
DO 5 J=IP1,10
5 T(ITO,J)=T(IFROM,J)+RM*PIV(J)
ITO=ITO-1
6 IFROM=IFROM-1
IF(IFROM.GE.I) GO TO 4
DO 7 J=IP1,10
7 T(I,J)=PIV(J)
DO 8 I=1,8
J=10-I
K=9-I
DO 8 L=J,9
8 T(K,10)=T(K,10)-T(K,L)*T(L,10)
DO 9 I=1,9
9 V(I)=T(I,10)
RETURN
END

```

```

SUBROUTINE OUTPUT
DIMENSION h(5,4),ANS1(96),ANS2(121),ANS3(56),ANS4(72)
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IGUMP ,IAMTP ,
2IGASMX,IOBURN,IAFTBN,IOCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOHAP,NUMHAP,MAPEOG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRI DS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETA CDS ,WACDS ,PRCCF ,ETAACF ,WACCF ,
5T4JS ,WF BDS ,DTCODS ,ETA BDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIFDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPDS ,CNLPDS ,ETL PDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WF ODS ,DTDU DS ,ETA DDS ,WA23DS ,DPODS ,DTDCUF ,ETAOCF ,
AT7DS ,WFA DS ,DTAFDS ,ETA ADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BL1 ,BLC ,BLDU ,
6CNF ,PRF ,ETA F ,WAF C ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETA I ,WAIC ,WAI ,BLOBI ,BLOB C ,WA3 ,
8CNC ,PRC ,ETA C ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,GHTI ,BLIP ,BLIPI ,BLIFC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CNG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DAL TP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FCBLDUC ,PCBL CBI ,PCBL OBC ,PCBLHPI ,PCBLHPC ,PCBLIFI ,PCBLIPC ,PCBL LPI ,
GPCBL LPC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XELF ,XBLOU ,XBLOUI ,XBLOUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6HAD ,WFD ,WG24 ,FAR24 ,ETAD ,DPOUC ,BYPASS ,DUMS7 ,
7TS29 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
EQUIVALENCE (ANS1(1),PCNFGU), (ANS2(1),T1), (ANS3(1),XP1)

```

```

EQUIVALENCE (ANS4(1),XT55)
DATA AWORD1,AWORD2/6HOUTPUT,6HCOMMON/
DATA (W(1,I),I=1,4)/6HSUBSON,6HIC C-D,6H NOZZL,6HE /
DATA (W(2,I),I=1,4)/6HSHOCK ,6HINSIDE,6H C-D N,6HOZZLE /
DATA (W(3,I),I=1,4)/6HSHOCK ,6HOUTSID,6HE C-D ,6HNOZZLE/
DATA (W(4,I),I=1,4)/6HSUBSON,6HIC CON,6HVERG. ,6HNOZZLE/
DATA (W(5,I),I=1,4)/6HSONIC ,6HCONVER,6HGENT N,6HOZZLE /
WORD=AWORD1
IF (IDES.EQ.1) GO TO 4
IF (IDBURN.GT.0) GO TO 2
IF (IAFTBN.GT.0) GO TO 1
WRITE (6,100) WORD,AM,ALTP,T4,ETAR
GO TO 3
1 WRITE (6,101) WORD,AM,ALTP,T4,T7,ETAR
GO TO 3
2 WRITE (6,102) WORD,AM,ALTP,T4,T24,ETAR
3 CALL CONOUT(2)
4 WRITE (6,104) (W(IMSHOC,I),I=1,4),FG,FN,SFC
IF (IGASHX.GT.0) GO TO 5
WRITE (6,105) (W(IDSHOC,I),I=1,4)
5 WRITE (6,106) LOOPER
WORD=AWORD2
WRITE (6,107) WORD,ZF,PCNF,ZI,PCNI,ZC,PCNC,T4,MODE
WRITE (6,108)
WRITE (6,109) (ANS1(I),I=1,96)
WRITE (6,108)
WRITE (6,109) (ANS2(I),I=1,121)
WRITE (6,108)
WRITE (6,109) (ANS3(I),I=1,56)
WRITE (6,108)
WRITE (6,109) (ANS4(I),I=1,72)
IF (IDES.EQ.1) GO TO 6
A8=A8SAV
A9=A9SAV
A28=A28SAV
A29=A29SAV
IF (IDUMP.NE.2) GO TO 6
WRITE (6,110)
CALL SYG(2)
6 CALL ENGBAL
RETURN
100 FORMAT(1H0,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F8.2,25X7H ETAR=,F7.4)
101 FORMAT(1H0,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F8.2,5X7H T7=,F8.2,5X7H ETAR=,F7.4)
102 FORMAT(1H0,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F9.2,5X7H T24=,F8.2,5X7H ETAR=,F7.4)
104 FORMAT(6H0MAIN ,4A6,9X3HFG=,F9.2,18X3HFN=,F9.2,18X4HSFC=,F8.5)
105 FORMAT(6H CUCT ,4A6)
106 FORMAT(16H1CCONVERGED AFTER,I4,6H LOOPS,/,1H0)
107 FORMAT(1H ,A6,9X,7E15.6,I4)
108 FORMAT(1H )
109 FORMAT(1H ,8E15.6)
110 FORMAT(1H1)
END

```

```

SUBROUTINE MAFBAC(MAP,MAPGO,TFFS,TFF,CNS,CN,PCN,T,MODE,IGO,NUM)
DATA WH,WI,WL,WT,WS/6H H.P. ,6H I.P. ,6H L.P. ,6H TFF ,6HSPEED /
1  FORMAT(1H0,A6,12HTURBINE MAP ,A6,4HWAS=,E13.6,10H AND NOW=,E13.6,
16H$$$$$$)
2  FORMAT(1H0,A6,A6,22HWAS ALSO CHANGED FROM ,E13.6,5H TO ,E13.6,
16H$$$$$$)
  IF(NUM.GT.0) GO TO 3
  NUMH=0
  NUMI=0
  NUML=0
3  IGO=MAPGO+3*(MAP-1)
  GO TO (100,200,300,400,500,500,700,800,900),IGO
C *** HIGH PRESSURE TURBINE
100 TFF=TFF+0.1*(TFF-TFFS)
  WRITE(8,1)WH,WT,TFFS,TFF
  RETURN
200 CN=CN+0.05*(CN-CNS)
  IF(MODE.NE.1) PCN=PCN*(CN/CNS)
  IF(MODE.EQ.1) T =T *(CNS/CN)**2
  WRITE(8,1)WH,WS,CNS,CN
  IF(NUMH.GT.2) GO TO 210
  NUMH=1
  NUMH=NUMH+1
  RETURN
210 DELCN=CN-CNS
  IF(DELCN.GE.0.) RETURN
  TFF=TFF*(1.+DELCN/CN)
  WRITE(8,2)WH,WT,TFFS,TFF
  RETURN
300 TFF=TFF+0.1*(TFF-TFFS)
  WRITE(8,1)WH,WT,TFFS,TFF
  GO TO 200
C *** LOW PRESSURE TURBINE
400 TFF=TFF+0.1*(TFF-TFFS)
  WRITE(8,1)WL,WT,TFFS,TFF
  RETURN
500 CN=CN+0.05*(CN-CNS)
  IF(MODE.NE.3) PCN=PCN*(CN/CNS)
  IF(MODE.EQ.3) T =T *(CNS/CN)
  WRITE(8,1)WL,WS,CNS,CN
  IF(NUML.GT.2) GO TO 510
  NUMI=1
  NUML=NUML+1
  RETURN
510 DELCN=CN-CNS
  IF(DELCN.GE.0.) RETURN
  TFF=TFF*(1.+DELCN/CN)
  WRITE(8,2)WL,WT,TFFS,TFF
  RETURN
600 TFF=TFF+0.1*(TFF-TFFS)
  WRITE(8,1)WL,WT,TFFS,TFF
  GO TO 500
C *** INTERMEDIATE SPOOL TURBINE
700 TFF=TFF+0.1*(TFF-TFFS)
  WRITE(8,1)WI,WT,TFFS,TFF

```

```

      RETURN
800  CN=CN+0.05*(CN-CNS)
      IF (MODE.NE.3) PCN=PCN*(CN/CNS)
      IF (MODE.EQ.3) T=T*(CNS/CN)
      WRITE (8,1) WI,WS,CNS,CN
      IF (NUMI.GT.2) GO TO 810
      NUM=1
      NUMI=NUMI+1
      RETURN
810  DELCN=CN-CNS
      IF (DELCN.GE.0.) RETURN
      TFF=TFF*(1.+DELCN/CN)
      WRITE (8,2) T,WT,TFFS,TFF
      RETURN
900  TFF=TFF+0.1*(TFF-TFFS)
      WRITE (8,1) WI,WT,TFFS,TFF
      GO TO 800
      END

```



```

SUBROUTINE THERMO(PX,HX,TX,SX,AMX,L,FAR,K)
FX=0.
IF (L.EQ.1) FX=FAR
IF (K.EQ.1) GO TO 1
CALL PROCOM(FX,TX,CS,AK,CP,R,PHI,HX)
GO TO 3
1 TX=4.*HX
DO 2 I=1,15
CALL FROCOM(FX,TX,CS,AK,CP,R,PHI,t)
DELH=HX-H
IF (ABS(DELH).LE.0.00001*HX) GO TO 3
2 TX=TX+4.*DELH
WRITE(8,100)
100 FORMAT(31HONG CONVERGENCE IN THERMO$$$)
3 SX=PHI-R*A LOG(PX)
AMX=1.986375/R
RETURN
END

```

```
SUBROUTINE RAM(AM,ETAR)
IF (AM.GT.1.) GO TO 2
ETAR=1.
1  RETURN
2  IF (AM.GT.5.) GO TO 3
   ETAR=1.-0.075*((AM-1.)**1.35)
   GO TO 1
3  ETAR=800./((AM**4)+935.)
   GO TO 1
END
```

```

FUNCTION GUESS(M,T,TD,P,PD,N,WD,C,DD,VD)
IF (M.EQ.0) GUESS=VD*((T/TD)**1.60)*((DD/D)**0.50)
IF (M.EQ.1) GUESS=VD*((P/PD)**1.80)*((DD/D)**0.33)
IF (M.EQ.2) GUESS=VD*((W/WD)**0.33)*((DD/D)**1.00)
IF (M.EQ.3) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)
IF (M.EQ.4) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)
IF (M.EQ.5) GUESS=VD*((T/TD)**1.10)*((DD/D)**0.60)
IF (M.EQ.6) GUESS=VD*((P/PD)**1.00)*((D/DD)**0.25)
IF (M.EQ.7) GUESS=VD*((P/PD)**0.62)*((D/DD)**0.31)
IF (M.EQ.8) GUESS=VD*((T/TD)**1.2)*DD/D
IF (M.EQ.9) GUESS=VD*P/PD*((D/DD)**1.5)
RETURN
END

```

```
BLOCK DATA CMEDAT
COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),N,NP(15)
DATA N,NP/15,15*15/
DATA PSI/4.9116,9.8232,14.735,19.646,24.558,29.470,34.381,
139.293,44.207,73.674,100.,200.,300.,400.,500./
DATA DELT/15*200.,15*300.,15*400.,15*500.,15*600.,15*700.,15*800.,
115*900.,15*1000.,15*1100.,15*1200.,15*1300.,15*1400.,15*1500.,
215*1600./
DATA ETA/
1.600,.726,.777,.806,.826,.843,.855,.865,7*.870,
2.758,.825,.858,.875,.888,.898,.906,.912,.914,6*.915,
3.868,.893,.911,.925,.935,.942,.947,.951,7*.953,
4.925,.936,.946,.955,.963,.969,.974,.977,.978,6*.979,
5.960,.966,.972,.977,.982,.985,.990,.992,.993,6*.995,
6.988,.991,.992,.994,.995,.997,.998,8*.999,
78*1.00,7*.999,120*1.00/
END
```

BLOCK DATA FANDAT

C THIS IS A GENERALIZED FAN MAP FOR UNREALISTIC SUPERSONIC ENGINE

COMMON/ FAN/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)

DATA N,NP/10,6,3*7,5*10,8,5*0/

DATA CN/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5*0./

DATA (PR(1,J) ,J=1,6)/

C1.0000 ,1.0120 ,1.0280 ,1.0384 ,1.0448 ,1.0480 /

DATA (WAC(1,J) ,J=1,6)/

C243.60 ,229.80 ,199.80 ,166.80 ,133.20 ,86.400 /

DATA (ETA(1,J) ,J=1,6)/

C0.7559 ,0.7612 ,0.7665 ,0.7559 ,0.7251 ,0.6415 /

DATA (PR(2,J) ,J=1,7)/

C1.0000 ,1.0200 ,1.0400 ,1.0584 ,1.0752 ,1.0920 ,1.1000 /

DATA (WAC(2,J) ,J=1,7)/

C286.80 ,270.00 ,253.20 ,233.40 ,209.40 ,183.60 ,156.60 /

DATA (ETA(2,J) ,J=1,7)/

C0.7559 ,0.7762 ,0.7920 ,0.7973 ,0.8026 ,0.7762 ,0.7401 /

DATA (PR(3,J) ,J=1,7)/

C1.0000 ,1.0256 ,1.0512 ,1.0800 ,1.1160 ,1.1320 ,1.1480 /

DATA (WAC(3,J) ,J=1,7)/

C333.60 ,322.80 ,310.20 ,291.60 ,259.80 ,240.00 ,213.60 /

DATA (ETA(3,J) ,J=1,7)/

C0.7506 ,0.7762 ,0.8026 ,0.8281 ,0.8439 ,0.8281 ,0.7662 /

DATA (PR(4,J) ,J=1,7)/

C1.0000 ,1.0368 ,1.0880 ,1.1240 ,1.1600 ,1.1896 ,1.1952 /

DATA (WAC(4,J) ,J=1,7)/

C383.40 ,376.20 ,358.20 ,340.20 ,313.20 ,276.60 ,266.40 /

DATA (ETA(4,J) ,J=1,7)/

C0.7454 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.8281 ,0.8078 /

DATA (PR(5,J) ,J=1,10)/

C1.0000 ,1.0640 ,1.1184 ,1.1480 ,1.1840 ,1.2096 ,1.2176 ,

C1.2240 ,1.2440 ,1.2672 /

DATA (WAC(5,J) ,J=1,10)/

C439.80 ,436.80 ,428.40 ,420.60 ,406.80 ,393.60 ,388.20 ,

C383.40 ,368.40 ,342.60 /

DATA (ETA(5,J) ,J=1,10)/

C0.7251 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9038 ,

C0.9011 ,0.8800 ,0.8281 /

DATA (PR(6,J) ,J=1,10)/

C1.0000 ,1.1000 ,1.1600 ,1.2000 ,1.2280 ,1.2552 ,1.2720 ,

C1.2864 ,1.3024 ,1.3320 /

DATA (WAC(6,J) ,J=1,10)/

C499.81 ,499.80 ,493.20 ,485.40 ,476.40 ,466.80 ,456.60 ,

C448.20 ,433.20 ,406.80 /

DATA (ETA(6,J) ,J=1,10)/

C0.6882 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9108 ,

C0.9011 ,0.8800 ,0.8272 /

DATA (PR(7,J) ,J=1,10)/

C1.0000 ,1.0760 ,1.1520 ,1.2192 ,1.2600 ,1.2896 ,1.3312 ,

C1.3616 ,1.3912 ,1.4000 /

DATA (WAC(7,J) ,J=1,10)/

C566.41 ,566.40 ,566.39 ,559.80 ,553.20 ,544.80 ,528.60 ,

C509.40 ,483.60 ,474.00 /

DATA (ETA(7,J) ,J=1,10)/

C0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,0.9011 ,

```

C0.8800 ,0.8281 ,0.8175 /
DATA (PR(8,J) ,J=1,10)/
C1.0000 ,1.0440 ,1.1352 ,1.2208 ,1.2944 ,1.3400 ,1.4000 ,
C1.4280 ,1.4480 ,1.4800 /
DATA (WAC(8,J) ,J=1,10)/
C633.61 ,633.60 ,633.59 ,633.00 ,625.80 ,616.80 ,600.00 ,
C586.80 ,576.60 ,553.20 /
DATA (ETA(8,J) ,J=1,10)/
C0.6002 ,0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,
C0.8589 ,0.8281 ,0.7867 /
DATA (PR(9,J) ,J=1,10)/
C1.0000 ,1.1040 ,1.2200 ,1.3240 ,1.4000 ,1.4480 ,1.5000 ,
C1.5336 ,1.5680 ,1.5840 /
DATA (WAC(9,J) ,J=1,10)/
C700.22 ,700.21 ,700.20 ,700.19 ,700.18 ,698.40 ,693.60 ,
C683.40 ,666.60 ,656.40 /
DATA (ETA(9,J) ,J=1,10)/
C0.5694 ,0.6415 ,0.7251 ,0.7762 ,0.8026 ,0.8078 ,0.8026 ,
C0.7762 ,0.7454 ,0.7251 /
DATA (PR(10,J) ,J=1,8 )/
C1.0000 ,1.1632 ,1.3120 ,1.4000 ,1.4800 ,1.5400 ,1.5800 ,
C1.6600 /
DATA (WAC(10,J) ,J=1,8 )/
C750.03 ,750.02 ,750.01 ,750.00 ,749.99 ,749.98 ,749.40 ,
C736.80 /
DATA (ETA(10,J) ,J=1,8 )/
C0.5174 ,0.6415 ,0.7251 ,0.7559 ,0.7612 ,0.7506 ,0.7251 ,
C0.6415 /
END

```

BLOCK DATA INTDAT
 THIS IS A GENERALIZED INT. COMP. MAP FOR UNREALISTIC SUPERSONIC EN
 COMMON/ INT/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)
 DATA N,NP/10,6,3*7,5*10,8,5*0/
 DATA CN/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5*0./
 DATA (PR(1,J) ,J=1,6)/
 C1.0000 ,1.0180 ,1.0420 ,1.0576 ,1.0672 ,1.0720 /
 DATA (WAC(1,J) ,J=1,6)/
 C121.80 ,114.90 ,99.900 ,83.400 ,66.600 ,43.200 /
 DATA (ETA(1,J) ,J=1,6)/
 C0.7559 ,0.7612 ,0.7665 ,0.7559 ,0.7251 ,0.6415 /
 DATA (PR(2,J) ,J=1,7)/
 C1.0000 ,1.0300 ,1.0600 ,1.0876 ,1.1128 ,1.1380 ,1.1500 /
 DATA (WAC(2,J) ,J=1,7)/
 C143.40 ,135.00 ,126.60 ,116.70 ,104.70 ,91.800 ,78.300 /
 DATA (ETA(2,J) ,J=1,7)/
 C0.7559 ,0.7762 ,0.7420 ,0.7973 ,0.8026 ,0.7762 ,0.7401 /
 DATA (PR(3,J) ,J=1,7)/
 C1.0000 ,1.0384 ,1.0768 ,1.1200 ,1.1740 ,1.1980 ,1.2220 /
 DATA (WAC(3,J) ,J=1,7)/
 C166.80 ,161.40 ,155.10 ,145.80 ,129.90 ,120.00 ,106.80 /
 DATA (ETA(3,J) ,J=1,7)/
 C0.7559 ,0.7762 ,0.8026 ,0.8281 ,0.8439 ,0.8281 ,0.7762 /
 DATA (PR(4,J) ,J=1,7)/
 C1.0000 ,1.0552 ,1.1320 ,1.1860 ,1.2400 ,1.2844 ,1.2928 /
 DATA (WAC(4,J) ,J=1,7)/
 C191.70 ,188.10 ,179.10 ,170.10 ,156.60 ,138.30 ,133.20 /
 DATA (ETA(4,J) ,J=1,7)/
 C0.7454 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.8281 ,0.8078 /
 DATA (PR(5,J) ,J=1,10)/
 C1.0000 ,1.0960 ,1.1776 ,1.2220 ,1.2760 ,1.3144 ,1.3264 ,
 C1.3360 ,1.3660 ,1.4008 /
 DATA (WAC(5,J) ,J=1,10)/
 C219.90 ,218.40 ,214.20 ,210.30 ,203.40 ,196.80 ,194.10 ,
 C191.70 ,184.20 ,171.30 /
 DATA (ETA(5,J) ,J=1,10)/
 C0.7251 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9038 ,
 C0.9011 ,0.8800 ,0.8281 /
 DATA (PR(6,J) ,J=1,10)/
 C1.0000 ,1.1500 ,1.2400 ,1.3000 ,1.3420 ,1.3828 ,1.4080 ,
 C1.4296 ,1.4536 ,1.4980 /
 DATA (WAC(6,J) ,J=1,10)/
 C249.91 ,249.90 ,246.60 ,242.70 ,238.20 ,233.40 ,228.30 ,
 C224.10 ,216.60 ,203.40 /
 DATA (ETA(6,J) ,J=1,10)/
 C0.6882 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9108 ,
 C0.9011 ,0.8800 ,0.8272 /
 DATA (PR(7,J) ,J=1,10)/
 C1.0000 ,1.1140 ,1.2280 ,1.3288 ,1.3900 ,1.4344 ,1.4968 ,
 C1.5424 ,1.5868 ,1.6000 /
 DATA (WAC(7,J) ,J=1,10)/
 C283.21 ,283.20 ,283.19 ,279.90 ,276.60 ,272.40 ,264.30 ,
 C254.70 ,241.80 ,237.00 /
 DATA (ETA(7,J) ,J=1,10)/
 C0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,0.9011 ,

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C0.8800 ,0.8281 ,0.8175 /
DATA (PR(8,J) ,J=1,10)/
C1.0000 ,1.0660 ,1.2028 ,1.3312 ,1.4416 ,1.5100 ,1.6000 ,
C1.6420 ,1.6720 ,1.7200 /
DATA (WAC(8,J) ,J=1,10)/
C316.81 ,316.80 ,316.79 ,316.50 ,312.90 ,308.40 ,300.00 ,
C293.40 ,288.30 ,276.60 /
DATA (ETA(8,J) ,J=1,10)/
C0.6002 ,0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,
C0.8589 ,0.8281 ,0.7867 /
DATA (PR(9,J) ,J=1,10)/
C1.0000 ,1.1560 ,1.3300 ,1.4860 ,1.6000 ,1.6720 ,1.7500 ,
C1.8004 ,1.8520 ,1.8760 /
DATA (WAC(9,J) ,J=1,10)/
C350.12 ,350.11 ,350.10 ,350.09 ,350.08 ,349.20 ,346.80 ,
C341.70 ,333.30 ,328.20 /
DATA (ETA(9,J) ,J=1,10)/
C0.5694 ,0.6415 ,0.7251 ,0.7762 ,0.8026 ,0.8078 ,0.8026 ,
C0.7762 ,0.7454 ,0.7251 /
DATA (PR(10,J) ,J=1,8 )/
C1.0000 ,1.2448 ,1.4680 ,1.6000 ,1.7200 ,1.8100 ,1.8700 ,
C1.9900 /
DATA (WAC(10,J) ,J=1,8 )/
C375.03 ,375.02 ,375.01 ,375.00 ,374.99 ,374.98 ,374.70 ,
C368.40 /
DATA (ETA(10,J) ,J=1,8 )/
C0.5174 ,0.6415 ,0.7251 ,0.7559 ,0.7612 ,0.7506 ,0.7251 ,
C0.6415 /
END

```


C

BLOCK DATA CHPOAT

THIS IS A GENERALIZED H.P. COMP. MAP FOR UNREALISTIC SUPERSONIC EN
COMMON/ COMP/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)

DATA N,NP/10,2*6,2*8,4*10,2*8,5*0/

DATA CN/.562,.674,.787,.899,1.,1.034,1.067,1.124,1.236,1.292,5*0./

DATA (PR(1,J) ,J=1,6)/

C1.0000 ,1.8400 ,2.4280 ,2.8690 ,3.8350 ,4.5490 /

DATA (WAC(1,J) ,J=1,6)/

C51.000 ,50.200 ,49.500 ,48.800 ,46.700 ,44.500 /

DATA (ETA(1,J) ,J=1,6)/

C0.5908 ,0.6218 ,0.6424 ,0.6527 ,0.6734 ,0.6424 /

DATA (PR(2,J) ,J=1,6)/

C1.0000 ,1.9660 ,3.0930 ,3.9330 ,4.6890 ,5.5290 /

DATA (WAC(2,J) ,J=1,6)/

C59.300 ,59.299 ,58.800 ,57.900 ,56.700 ,55.000 /

DATA (ETA(2,J) ,J=1,5)/

C0.5908 ,0.6424 ,0.6940 ,0.7250 ,0.7456 ,0.7250 /

DATA (PR(3,J) ,J=1,8)/

C1.0000 ,1.8400 ,2.6800 ,3.4080 ,4.5210 ,5.4450 ,6.3130 ,

C6.5230 /

DATA (WAC(3,J) ,J=1,8)/

C70.000 ,70.000 ,69.999 ,69.500 ,68.800 ,67.900 ,66.400 ,

C65.700 /

DATA (ETA(3,J) ,J=1,8)/

C0.5857 ,0.6424 ,0.6837 ,0.7250 ,0.7774 ,0.7929 ,0.7774 ,

C0.7697 /

DATA (PR(4,J) ,J=1,8)/

C1.0000 ,2.0080 ,3.4290 ,4.6050 ,5.6970 ,6.6140 ,7.5380 ,

C7.9580 /

DATA (WAC(4,J) ,J=1,8)/

C84.802 ,84.801 ,84.800 ,84.799 ,84.000 ,83.300 ,81.700 ,

C80.500 /

DATA (ETA(4,J) ,J=1,8)/

C0.5805 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8290 ,0.804 ,

C0.7929 /

DATA (PR(5,J) ,J=1,10)/

C1.0000 ,2.5190 ,3.9820 ,5.2770 ,6.4880 ,7.2020 ,8.0000 ,

C8.5670 ,9.3860 ,9.5960 /

DATA (WAC(5,J) ,J=1,10)/

C101.72 ,101.71 ,101.70 ,101.69 ,101.20 ,101.00 ,100.00 ,

C99.500 ,98.100 ,97.400 /

DATA (ETA(5,J) ,J=1,10)/

C0.5719 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8394 ,0.8600 ,

C0.8394 ,0.8084 ,0.8058 /

DATA (PR(6,J) ,J=1,10)/

C1.0000 ,2.8550 ,4.2970 ,5.6130 ,6.9360 ,7.6220 ,8.5460 ,

C9.1340 ,9.9250 ,10.2190 /

DATA (WAC(6,J) ,J=1,10)/

C108.12 ,108.11 ,108.10 ,108.09 ,107.60 ,107.10 ,106.70 ,

C106.00 ,104.50 ,104.00 /

DATA (ETA(6,J) ,J=1,10)/

C0.5702 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8394 ,0.8600 ,

C0.8394 ,0.8084 ,0.8041 /

DATA (PR(7,J) ,J=1,10)/

C1.0000 ,3.2610 ,4.7590 ,6.1170 ,7.4540 ,8.3080 ,9.2180 ,

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C9.6380 ,10.513 ,10.996 /
  DATA (WAC(7,J) ,J=1,10)/
C114.52 ,114.51 ,114.50 ,114.49 ,114.48 ,114.30 ,113.60 ,
C113.30 ,112.60 ,112.40 /
  DATA (ETA(7,J) ,J=1,10)/
C0.5599 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8394 ,0.8497 ,
C0.8394 ,0.8084 ,0.7981 /
  DATA (PR(8,J) ,J=1,10)/
C1.0000 ,1.6860 ,3.8490 ,5.4660 ,6.8660 ,8.3710 ,8.9660 ,
C9.8830 ,10.912 ,11.815 /
  DATA (WAC(8,J) ,J=1,10)/
C122.93 ,122.92 ,122.91 ,122.90 ,122.89 ,122.88 ,122.60 ,
C122.10 ,121.70 ,120.70 /
  DATA (ETA(8,J) ,J=1,10)/
C0.5392 ,0.5702 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8239 ,
C0.8394 ,0.8084 ,0.7774 /
  DATA (PR(9,J) ,J=1,8 )/
C1.0000 ,4.3530 ,7.6220 ,10.219 ,11.059 ,11.899 ,13.159 ,
C13.656 /
  DATA (WAC(9,J) ,J=1,8 )/
C139.82 ,139.81 ,139.80 ,139.79 ,139.78 ,139.50 ,139.30 ,
C139.00 /
  DATA (ETA(9,J) ,J=1,8 )/
C0.4764 ,0.6011 ,0.7250 ,0.7774 ,0.7826 ,0.7774 ,0.7250 ,
C0.6992 /
  DATA (PR(10,J) ,J=1,8 )/
C1.0000 ,3.7650 ,6.4810 ,9.1760 ,10.219 ,11.479,12.711 ,
C14.412 /
  DATA (WAC(10,J) ,J=1,8 )/
C146.24 ,146.23 ,146.22 ,146.21 ,146.20 ,146.19 ,146.18 ,
C146.17 /
  DATA (ETA(10,J) ,J=1,8 )/
C0.4661 ,0.5702 ,0.6424 ,0.7250 ,0.7508 ,0.7508 ,0.7250 ,
C0.6424 /
  END

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```

BLOCK DATA LTURB
C THIS IS A GENERALIZED L.P. TURBINE MAP
COMMON/ LTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)
DATA N,NP/11,9*15,12,9,4*0/
DATA TFF/ 88.470, 102.80, 116.84, 129.33, 141.05, 145.73, 150.00,
1 153.35, 156.41, 159.78, 163.17, 4*0./
DATA ( CN( 1,J),J=1,15)/
C0.3682, 0.5336, 0.7365, 0.9754, 1.2146, 1.4173, 1.6201, 1.7673,
C2.0247, 2.2827, 2.4665, 2.6137, 2.8166, 2.9456, 3.3138/
DATA ( DH( 1,J),J=1,15)/
C0.0010, 0.0026, 0.0035, 0.0044, 0.0051, 0.0056, 0.0059, 0.0061,
C0.0062, 0.0061, 0.0057, 0.0053, 0.0044, 0.0035, 0.0001/
DATA (ETA( 1,J),J=1,15)/
C0.7120, 0.7300, 0.7472, 0.7300, 0.7140, 0.7008, 0.6850, 0.6730,
C0.6452, 0.6200, 0.6000, 0.5750, 0.5310, 0.5000, 0.3850/
DATA ( CN( 2,J),J=1,15)/
C0.3682, 0.5518, 0.7919, 1.0672, 1.2882, 1.4446, 1.6937, 1.8954,
C2.0619, 2.2273, 2.3747, 2.6229, 2.8720, 3.0555, 3.3138/
DATA ( DH( 2,J),J=1,15)/
C0.0026, 0.0039, 0.0054, 0.0069, 0.0080, 0.0087, 0.0096, 0.0101,
C0.0104, 0.0107, 0.0108, 0.0106, 0.0101, 0.0094, 0.0077/
DATA (ETA( 2,J),J=1,15)/
C0.8000, 0.8100, 0.8200, 0.8300, 0.8300, 0.8298, 0.8100, 0.8000,
C0.7850, 0.7600, 0.7450, 0.7000, 0.6800, 0.6450, 0.5900/
DATA ( CN( 3,J),J=1,15)/
C0.3682, 0.5911, 0.8655, 1.0764, 1.2519, 1.4354, 1.6201, 1.8409,
C2.0247, 2.2455, 2.4302, 2.5956, 2.7791, 3.0555, 3.3138/
DATA ( DH( 3,J),J=1,15)/
C0.0031, 0.0051, 0.0071, 0.0087, 0.0099, 0.0111, 0.0122, 0.0134,
C0.0143, 0.0152, 0.0157, 0.0162, 0.0166, 0.0167, 0.0164/
DATA (ETA( 3,J),J=1,15)/
C0.8000, 0.8300, 0.8600, 0.8630, 0.8670, 0.8700, 0.8720, 0.8720,
C0.8700, 0.8670, 0.8600, 0.8500, 0.8300, 0.8000, 0.7600/
DATA ( CN( 4,J),J=1,15)/
C0.3682, 0.4237, 0.6810, 0.8837, 1.1047, 1.2882, 1.5090, 1.7482,
C2.0429, 2.2091, 2.3747, 2.6047, 2.8720, 3.1291, 3.3138/
DATA ( DH( 4,J),J=1,15)/
C0.0033, 0.0038, 0.0061, 0.0078, 0.0096, 0.0110, 0.0126, 0.0141,
C0.0159, 0.0166, 0.0174, 0.0183, 0.0191, 0.0195, 0.0197/
DATA (ETA( 4,J),J=1,15)/
C0.7995, 0.8000, 0.8400, 0.8600, 0.8680, 0.8730, 0.8800, 0.8830,
C0.8835, 0.8830, 0.8800, 0.8740, 0.8600, 0.8350, 0.8200/
DATA ( CN( 5,J),J=1,15)/
C0.3682, 0.5065, 0.7365, 0.9754, 1.2882, 1.5647, 1.7301, 1.9690,
C2.0983, 2.2637, 2.4332, 2.6691, 2.9456, 3.1846, 3.3138/
DATA ( DH( 5,J),J=1,15)/
C0.0036, 0.0049, 0.0071, 0.0092, 0.0119, 0.0141, 0.0155, 0.0172,
C0.0181, 0.0192, 0.0202, 0.0214, 0.0226, 0.0235, 0.0239/
DATA (ETA( 5,J),J=1,15)/
C0.7750, 0.8000, 0.8480, 0.8600, 0.8750, 0.8900, 0.8912, 0.8940,
C0.8955, 0.8970, 0.8961, 0.8900, 0.8790, 0.8671, 0.8600/
DATA ( CN( 6,J),J=1,15)/
C0.3682, 0.6164, 0.8372, 1.1047, 1.2882, 1.5283, 1.7482, 1.9509,
C2.2133, 2.4302, 2.6510, 2.8619, 3.1384, 3.2584, 3.3138/
DATA ( DH( 6,J),J=1,15)/

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C0.0038, 0.0064, 0.0087, 0.0113, 0.0130, 0.0152, 0.0171, 0.0187,
 C0.0209, 0.0226, 0.0244, 0.0259, 0.0286, 0.0303, 0.0319/
 DATA (ETA(6,J),J=1,15)/
 C0.7600, 0.8000, 0.8450, 0.8600, 0.8730, 0.8900, 0.8950, 0.9000,
 C0.9005, 0.9010, 0.9004, 0.9000, 0.8900, 0.8800, 0.8735/
 DATA (CN(7,J),J=1,15)/
 C0.3682, 0.7728, 1.0129, 1.2659, 1.4729, 1.6785, 1.8409, 2.0247,
 C2.1901, 2.3000, 2.3929, 2.5038, 2.5583, 2.6137, 2.6319/
 DATA (DH(7,J),J=1,15)/
 C0.0044, 0.0089, 0.0115, 0.0141, 0.0162, 0.0181, 0.0197, 0.0216,
 C0.0235, 0.0250, 0.0265, 0.0284, 0.0296, 0.0314, 0.0329/
 DATA (ETA(7,J),J=1,15)/
 C0.7310, 0.8000, 0.8300, 0.8600, 0.8750, 0.8900, 0.8930, 0.8975,
 C0.8999, 0.9000, 0.8980, 0.8937, 0.8900, 0.8799, 0.8710/
 DATA (CN(8,J),J=1,15)/
 C0.3682, 0.6072, 0.7919, 0.9754, 1.2337, 1.4548, 1.6383, 1.8409,
 C1.9509, 2.0801, 2.1537, 2.2091, 2.2637, 2.3009, 2.3051/
 DATA (DH(8,J),J=1,15)/
 C0.0048, 0.0078, 0.0102, 0.0124, 0.0153, 0.0177, 0.0201, 0.0226,
 C0.0242, 0.0261, 0.0274, 0.0285, 0.0299, 0.0314, 0.0321/
 DATA (ETA(8,J),J=1,15)/
 C0.7100, 0.7450, 0.7680, 0.8000, 0.8380, 0.8600, 0.8712, 0.8780,
 C0.8800, 0.8775, 0.8760, 0.8722, 0.8660, 0.8600, 0.8480/
 DATA (CN(9,J),J=1,15)/
 C0.3682, 0.5518, 0.6629, 0.8282, 1.0129, 1.1691, 1.2337, 1.3809,
 C1.5283, 1.6201, 1.7482, 1.8409, 1.8954, 1.9147, 1.9237/
 DATA (DH(9,J),J=1,15)/
 C0.0054, 0.0080, 0.0096, 0.0119, 0.0141, 0.0160, 0.0169, 0.0188,
 C0.0209, 0.0223, 0.0244, 0.0263, 0.0279, 0.0289, 0.0303/
 DATA (ETA(9,J),J=1,15)/
 C0.6780, 0.7000, 0.7125, 0.7350, 0.7690, 0.8000, 0.8060, 0.8225,
 C0.8395, 0.8450, 0.8470, 0.8445, 0.8330, 0.8235, 0.8080/
 DATA (CN(10,J),J=1,12)/
 C0.3682, 0.4782, 0.6447, 0.7546, 0.8655, 0.9754, 1.1047, 1.2015,
 C1.2701, 1.3073, 1.3365, 1.3407/
 DATA (DH(10,J),J=1,12)/
 C0.0061, 0.0078, 0.0104, 0.0122, 0.0139, 0.0157, 0.0181, 0.0201,
 C0.0217, 0.0230, 0.0244, 0.0251/
 DATA (ETA(10,J),J=1,12)/
 C0.6380, 0.6550, 0.6700, 0.6850, 0.7000, 0.7110, 0.7180, 0.7180,
 C0.7170, 0.7140, 0.7000, 0.6890/
 DATA (CN(11,J),J=1,9)/
 C0.3682, 0.4418, 0.5518, 0.6447, 0.7365, 0.8282, 0.8837, 0.9391,
 C0.9715/
 DATA (DH(11,J),J=1,9)/
 C0.0069, 0.0086, 0.0106, 0.0123, 0.0141, 0.0159, 0.0172, 0.0186,
 C0.0201/
 DATA (ETA(11,J),J=1,9)/
 C0.6000, 0.6000, 0.6120, 0.6170, 0.6210, 0.6258, 0.6250, 0.6230,
 C0.6089/
 END

```

BLOCK DATA ITURB
C THIS IS A GENERALIZED I.P. TURBINE MAP
COMMON/ ITURE/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)
DATA N,NP/11,9*15,12,9,4*0/
DATA TFF/ 39.670, 82.236, 93.468, 103.464, 112.836, 116.580,
1 120.000, 122.68, 125.12, 127.82, 130.536, 4*0./
DATA ( CN( 1,J),J=1,15)/
C0.3522, 0.5194, 0.7044, 0.9330, 1.1618, 1.3556, 1.5497, 1.6905,
C1.9367, 2.1835, 2.3593, 2.5001, 2.6941, 2.8175, 3.1698/
DATA ( DH( 1,J),J=1,15)/
C0.0016, 0.0023, 0.0031, 0.0038, 0.0045, 0.0049, 0.0052, 0.0054,
C0.0055, 0.0054, 0.0051, 0.0047, 0.0038, 0.0031, 0.0001/
DATA (ETA( 1,J),J=1,15)/
C0.7120, 0.7300, 0.7472, 0.7300, 0.7140, 0.7000, 0.6850, 0.6730,
C0.6452, 0.6200, 0.6000, 0.5750, 0.5310, 0.5000, 0.3850/
DATA ( CN( 2,J),J=1,15)/
C0.3522, 0.5278, 0.7575, 1.0208, 1.2322, 1.3818, 1.6201, 1.8130,
C1.9723, 2.1305, 2.2715, 2.5089, 2.7471, 2.9227, 3.1698/
DATA ( DH( 2,J),J=1,15)/
C0.0023, 0.0035, 0.0047, 0.0061, 0.0070, 0.0076, 0.0084, 0.0089,
C0.0092, 0.0094, 0.0095, 0.0093, 0.0089, 0.0083, 0.0068/
DATA (ETA( 2,J),J=1,15)/
C0.8000, 0.8100, 0.8200, 0.8300, 0.8300, 0.8290, 0.8100, 0.8000,
C0.7800, 0.7600, 0.7450, 0.7000, 0.6800, 0.6450, 0.5900/
DATA ( CN( 2,J),J=1,15)/
C0.3522, 0.5654, 0.8279, 1.0296, 1.1975, 1.3730, 1.5497, 1.7609,
C1.9367, 2.1479, 2.3245, 2.4827, 2.6583, 2.9227, 3.1698/
DATA ( DH( 2,J),J=1,15)/
C0.0027, 0.0045, 0.0063, 0.0076, 0.0087, 0.0098, 0.0107, 0.0118,
C0.0126, 0.0134, 0.0139, 0.0142, 0.0146, 0.0147, 0.0145/
DATA (ETA( 2,J),J=1,15)/
C0.8000, 0.8300, 0.8600, 0.8630, 0.8670, 0.8700, 0.8720, 0.8720,
C0.8700, 0.8670, 0.8600, 0.8500, 0.8300, 0.8000, 0.7600/
DATA ( CN( 3,J),J=1,15)/
C0.3522, 0.5654, 0.8279, 1.0296, 1.1975, 1.3730, 1.5497, 1.7609,
C1.9367, 2.1479, 2.3245, 2.4827, 2.6583, 2.9227, 3.1698/
DATA ( DH( 3,J),J=1,15)/
C0.0027, 0.0045, 0.0063, 0.0076, 0.0087, 0.0098, 0.0107, 0.0118,
C0.0126, 0.0134, 0.0139, 0.0142, 0.0146, 0.0147, 0.0145/
DATA (ETA( 3,J),J=1,15)/
C0.8000, 0.8300, 0.8600, 0.8630, 0.8670, 0.8700, 0.8720, 0.8720,
C0.8700, 0.8670, 0.8600, 0.8500, 0.8300, 0.8000, 0.7600/
DATA ( CN( 4,J),J=1,15)/
C0.3522, 0.4052, 0.6514, 0.8452, 1.0567, 1.2322, 1.4434, 1.6722,
C1.9540, 2.1131, 2.2715, 2.4915, 2.7471, 2.9931, 3.1698/
DATA ( DH( 4,J),J=1,15)/
C0.0029, 0.0034, 0.0054, 0.0069, 0.0084, 0.0097, 0.0111, 0.0124,
C0.0140, 0.0146, 0.0153, 0.0161, 0.0168, 0.0172, 0.0173/
DATA (ETA( 4,J),J=1,15)/
C0.7995, 0.8000, 0.8400, 0.8600, 0.8680, 0.8730, 0.8800, 0.8830,
C0.8835, 0.8830, 0.8800, 0.8740, 0.8600, 0.8350, 0.8200/
DATA ( CN( 5,J),J=1,15)/
C0.3522, 0.4844, 0.7044, 0.9330, 1.2322, 1.4967, 1.6548, 1.8834,
C2.0071, 2.1652, 2.3274, 2.5531, 2.8175, 3.0461, 3.1698/
DATA ( DH( 5,J),J=1,15)/

```

C0.0031, 0.0043, 0.0062, 0.0081, 0.0105, 0.0124, 0.0136, 0.0152,
 C0.0159, 0.0169, 0.0178, 0.0189, 0.0199, 0.0207, 0.0210/
 DATA (ETA(5,J),J=1,15)/
 C0.7750, 0.8000, 0.8480, 0.8600, 0.8750, 0.8900, 0.8912, 0.8940,
 C0.8955, 0.8970, 0.8961, 0.8900, 0.8790, 0.8671, 0.8600/
 DATA (CN(6,J),J=1,15)/
 C0.3522, 0.5896, 0.8008, 1.0567, 1.2322, 1.4619, 1.6722, 1.8660,
 C2.1171, 2.3245, 2.5357, 2.7375, 3.0019, 3.1167, 3.1698/
 DATA (DH(6,J),J=1,15)/
 C0.0034, 0.0057, 0.0076, 0.0100, 0.0114, 0.0134, 0.0150, 0.0165,
 C0.0184, 0.0199, 0.0214, 0.0228, 0.0251, 0.0267, 0.0290/
 DATA (ETA(6,J),J=1,15)/
 C0.7600, 0.8000, 0.8150, 0.8500, 0.8730, 0.8900, 0.8950, 0.9000,
 C0.9005, 0.9010, 0.9004, 0.9000, 0.8900, 0.8800, 0.8710/
 DATA (CN(7,J),J=1,15)/
 C0.3522, 0.7392, 0.9689, 1.2109, 1.4089, 1.6056, 1.7609, 1.9367,
 C2.0948, 2.2000, 2.2889, 2.3949, 2.4471, 2.5001, 2.5175/
 DATA (DH(7,J),J=1,15)/
 C0.0038, 0.0078, 0.0101, 0.0124, 0.0142, 0.0159, 0.0173, 0.0190,
 C0.0207, 0.0220, 0.0233, 0.0250, 0.0261, 0.0276, 0.0290/
 DATA (ETA(7,J),J=1,15)/
 C0.7310, 0.8000, 0.8300, 0.8600, 0.8750, 0.8900, 0.8930, 0.8975,
 C0.8999, 0.9000, 0.8980, 0.8937, 0.8900, 0.8799, 0.8710/
 DATA (CN(8,J),J=1,15)/
 C0.3522, 0.5808, 0.7575, 0.9330, 1.1801, 1.3915, 1.5671, 1.7609,
 C1.8660, 1.9897, 2.0601, 2.1131, 2.1652, 2.2009, 2.2048/
 DATA (DH(8,J),J=1,15)/
 C0.0042, 0.0069, 0.0090, 0.0109, 0.0135, 0.0156, 0.0177, 0.0199,
 C0.0213, 0.0230, 0.0241, 0.0251, 0.0263, 0.0276, 0.0283/
 DATA (ETA(8,J),J=1,15)/
 C0.7100, 0.7450, 0.7680, 0.8000, 0.8380, 0.8600, 0.8712, 0.8780,
 C0.8800, 0.8775, 0.8760, 0.8722, 0.8660, 0.8600, 0.8480/
 DATA (CN(9,J),J=1,15)/
 C0.3522, 0.5278, 0.6340, 0.7422, 0.9689, 1.1183, 1.1801, 1.3209,
 C1.4619, 1.5497, 1.6722, 1.7609, 1.8130, 1.8315, 1.8481/
 DATA (DH(9,J),J=1,15)/
 C0.0047, 0.0070, 0.0084, 0.0104, 0.0124, 0.0141, 0.0148, 0.0166,
 C0.0184, 0.0196, 0.0214, 0.0232, 0.0245, 0.0255, 0.0267/
 DATA (ETA(9,J),J=1,15)/
 C0.6780, 0.7000, 0.7125, 0.7350, 0.7690, 0.8000, 0.8060, 0.8225,
 C0.8395, 0.8450, 0.8470, 0.8445, 0.8330, 0.8235, 0.8080/
 DATA (CN(10,J),J=1,12)/
 C0.3522, 0.4574, 0.6167, 0.7218, 0.8279, 0.9330, 1.0567, 1.1493,
 C1.2148, 1.2505, 1.2784, 1.2824/
 DATA (DH(10,J),J=1,12)/
 C0.0054, 0.0069, 0.0092, 0.0107, 0.0123, 0.0138, 0.0159, 0.0177,
 C0.0191, 0.0202, 0.0214, 0.0221/
 DATA (ETA(10,J),J=1,12)/
 C0.6380, 0.6550, 0.6700, 0.6850, 0.7000, 0.7110, 0.7180, 0.7180,
 C0.7170, 0.7140, 0.7000, 0.6890/
 DATA (CN(11,J),J=1,9)/
 C0.3522, 0.4226, 0.5278, 0.6167, 0.7046, 0.7422, 0.8452, 0.8983,
 C0.9293/
 DATA (DH(11,J),J=1,9)/
 C0.0061, 0.0075, 0.0093, 0.0108, 0.0124, 0.0140, 0.0151, 0.0164,

```
CO.0177/  
DATA (ETA(11,J),J=1,9 )/  
CO.6000; 0.6000, 0.6120, 0.6170, 0.6210, 0.6258, 0.6250, 0.6230,  
CO.6009/  
END
```

C

BLOCK DATA HTURB

THIS IS A GENERALIZED H.P. TURBINE MAP

COMMON / HTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)

DATA N,NP/10,9*15,12,5*0/

DATA TFF/ 39.670, 42.990, 47.460, 48.610, 49.175, 49.600,
1 50.000, 50.425, 50.920, 51.575, 5*0./

DATA (CN(1,J),J=1,15)/

C0.1872, 0.3372, 0.5156, 0.7128, 0.9382, 1.1442, 1.3138, 1.5382,
C1.7264, 1.9324, 2.1500, 2.4058, 2.5892, 2.7862, 2.9460/

DATA (DH(1,J),J=1,15)/

C0.0032, 0.0057, 0.0084, 0.0108, 0.0133, 0.0152, 0.0164, 0.0174,
C0.0179, 0.0176, 0.0167, 0.0144, 0.0120, 0.0082, 0.0034/

DATA (ETA(1,J),J=1,15)/

C0.6219, 0.7078, 0.7868, 0.8090, 0.8090, 0.7963, 0.7779, 0.7422,
C0.7078, 0.7635, 0.6058, 0.5309, 0.4773, 0.4045, 0.3034/

DATA (CN(2,J),J=1,15)/

C0.1872, 0.3942, 0.5814, 0.7128, 0.8442, 0.9804, 1.1068, 1.2754,
C1.4450, 1.7068, 1.9696, 2.2706, 2.6970, 3.0960, 3.3774/

DATA (DH(2,J),J=1,15)/

C0.0036, 0.0080, 0.0113, 0.0136, 0.0156, 0.0175, 0.0192, 0.0212,
C0.0228, 0.0248, 0.0260, 0.0261, 0.0241, 0.0188, 0.0128/

DATA (ETA(2,J),J=1,15)/

C0.6068, 0.7078, 0.8090, 0.8292, 0.8363, 0.8393, 0.8368, 0.8302,
C0.8254, 0.8090, 0.7696, 0.7078, 0.6066, 0.5056, 0.4197/

DATA (CN(3,J),J=1,15)/

C0.1872, 0.4362, 0.6568, 0.8726, 1.0696, 1.2382, 1.4638, 1.6882,
C1.9696, 2.2138, 2.5520, 2.8650, 3.0392, 3.2348, 3.3774/

DATA (DH(3,J),J=1,15)/

C0.0046, 0.0100, 0.0144, 0.0184, 0.0216, 0.0240, 0.0268, 0.0292,
C0.0316, 0.0331, 0.0344, 0.0346, 0.0340, 0.0324, 0.0312/

DATA (ETA(3,J),J=1,15)/

C0.5764, 0.7078, 0.8090, 0.8494, 0.8543, 0.8515, 0.8494, 0.8409,
C0.8262, 0.8090, 0.7579, 0.7078, 0.6652, 0.6068, 0.5865/

DATA (CN(4,J),J=1,15)/

C0.1872, 0.2550, 0.4784, 0.6942, 0.9148, 1.1442, 1.3862, 1.5618,
C1.8010, 1.9794, 2.2794, 2.5138, 2.8334, 3.1422, 3.3774/

DATA (DH(4,J),J=1,15)/

C0.0052, 0.0068, 0.0120, 0.0164, 0.0204, 0.0244, 0.0280, 0.0304,
C0.0336, 0.0356, 0.0388, 0.0412, 0.0441, 0.0472, 0.0494/

DATA (ETA(4,J),J=1,15)/

C0.5643, 0.6068, 0.7078, 0.8090, 0.8494, 0.8596, 0.8596, 0.8575,
C0.8535, 0.9494, 0.8363, 0.8262, 0.8090, 0.7797, 0.7584/

DATA (CN(5,J),J=1,15)/

C0.1872, 0.5000, 0.5254, 0.7500, 0.9754, 1.2754, 1.4824, 1.7638,
C2.9450, 2.3362, 2.6450, 2.8700, 3.0754, 3.1520, 3.1618/

DATA (DH(5,J),J=1,15)/

C0.0056, 0.0088, 0.0144, 0.0192, 0.0236, 0.0288, 0.0321, 0.0360,
C0.0400, 0.0444, 0.0496, 0.0540, 0.0596, 0.0640, 0.0661/

DATA (ETA(5,J),J=1,15)/

C0.5562, 0.6068, 0.7078, 0.8090, 0.8494, 0.8697, 0.8696, 0.8662,
C0.8615, 0.8559, 0.8520, 0.8494, 0.8494, 0.8532, 0.8570/

DATA (CN(6,J),J=1,15)/

C0.1872, 0.3568, 0.6196, 0.8628, 1.0932, 1.2052, 1.5010, 1.6882,
C1.9138, 2.1246, 2.2706, 2.4226, 2.4950, 2.5372, 2.5558/

DATA (DH(6,J),J=1,15)/

C0.0068, 0.0120, 0.0192, 0.0252, 0.0300, 0.0340, 0.0384, 0.0421,
 C0.0472, 0.0524, 0.0564, 0.0668, 0.0640, 0.0668, 0.0698/
 DATA (ETA(6,J),J=1,15)/
 C0.5309, 0.6068, 0.7078, 0.8090, 0.8494, 0.8697, 0.8819, 0.8899,
 C0.8940, 0.8969, 0.8975, 0.8937, 0.8968, 0.8937, 0.8896/
 DATA (CN(7,J),J=1,15)/
 C0.1872, 0.4314, 0.6844, 0.9568, 1.2010, 1.3834, 1.5108, 1.6186,
 C1.7450, 1.8618, 1.9558, 2.0000, 2.0450, 2.0824, 2.1010/
 DATA (DH(7,J),J=1,15)/
 C0.0080, 0.0164, 0.0236, 0.0308, 0.0372, 0.0416, 0.0448, 0.0476,
 C0.0510, 0.0544, 0.0576, 0.0600, 0.0624, 0.0660, 0.0700/
 DATA (ETA(7,J),J=1,15)/
 C0.5062, 0.6068, 0.7078, 0.8090, 0.8494, 0.8697, 0.8797, 0.8899,
 C0.8954, 0.9000, 0.9010, 0.9000, 0.8980, 0.8925, 0.8793/
 DATA (CN(8,J),J=1,15)/
 C0.1872, 0.4834, 0.7314, 0.8814, 1.0226, 1.1442, 1.2804, 1.3696,
 C1.4638, 1.5950, 1.6746, 1.7450, 1.8010, 1.8156, 1.8196/
 DATA (DH(8,J),J=1,15)/
 C0.0088, 0.0196, 0.0272, 0.0316, 0.0356, 0.0392, 0.0432, 0.0460,
 C0.0488, 0.0528, 0.0560, 0.0596, 0.0640, 0.0664, 0.0693/
 DATA (ETA(8,J),J=1,15)/
 C0.5051, 0.6068, 0.7078, 0.7665, 0.8090, 0.8292, 0.8494, 0.8596,
 C0.8697, 0.8808, 0.8848, 0.8848, 0.8788, 0.8697, 0.8590/
 DATA (CN(9,J),J=1,15)/
 C0.1872, 0.3372, 0.5344, 0.6754, 0.8068, 0.9196, 1.0128, 1.1254,
 C1.2196, 1.3138, 1.3696, 1.4068, 1.4450, 1.4638, 1.4676/
 DATA (DH(9,J),J=1,15)/
 C0.0093, 0.0159, 0.0232, 0.0284, 0.0330, 0.0388, 0.0400, 0.0442,
 C0.0480, 0.0524, 0.0556, 0.0580, 0.0612, 0.0648, 0.0668/
 DATA (ETA(9,J),J=1,15)/
 C0.4909, 0.5380, 0.6068, 0.6573, 0.7078, 0.7463, 0.7776, 0.8090,
 C0.8191, 0.8302, 0.8347, 0.8363, 0.8322, 0.8241, 0.8090/
 DATA (CN(10,J),J=1,12)/
 C0.1872, 0.2814, 0.3804, 0.4686, 0.5628, 0.6382, 0.6892, 0.7362,
 C0.7696, 0.8068, 0.8254, 0.8304/
 DATA (DH(10,J),J=1,12)/
 C0.0132, 0.0180, 0.0228, 0.0268, 0.0314, 0.0352, 0.0380, 0.0412,
 C0.0440, 0.0476, 0.0504, 0.0530/
 DATA (ETA(10,J),J=1,12)/
 C0.4257, 0.4747, 0.5056, 0.5359, 0.5683, 0.5941, 0.6168, 0.6178,
 C0.6240, 0.6310, 0.6265, 0.6118/
 END

SECTION VIII
SAMPLE INPUT LISTING

The first section of data cards is the set of variables selected for output (controlled output). The second section is the Namelist input for running the desired points. The Namelist input consists of the following points: design point at sea-level static, a setup case for sea-level static afterburning, a sea-level static full afterburning point, several points in a subsonic power hook, a setup case for supersonic afterburning, and the supersonic full afterburning point.

T75
P55
T24
P24
WAD
WFO
WG24
FAR24
T25
P25
DPDUC
ETAD
AM2F
AM55
ETAF
ETAI
ETAC
ETATHC
ETATIF
ETATLC
T6
P6
A46
V6
WG6
T7
WFA
FAR7
ETAA
DPAFT
PSR
A48
V8
PS9
A49
V9
PS2A
AM2A
V28
PS2C
AM2C
V29
BYEASS
HPEXT
WFT
WGT
VA
FRD
CVMN07
VJM
CVQ007
VJD
FGM
FGP

A55
A75
A6
A8
A9
A28
A29

THEEND

\$DATAIN ITITLE=1,IDES=1,IDUMP=1,IAMTP=0,MODE=3,
IGASMY=2,TMCD=1,PS55=2.0,AM55=0.00,AM6=0.24,MPEXT=0.,NCZFLT=0,
ITRYS=200.,TOLALL=.005,ZFDS=0.8333,ZIDS=0.8143,ZGDS=0.8143,
PCNFDS=100.,ETAFDS=0.839,PCNIDS=100.,ETAIDS=0.853,PCNCPS=100.,
ETACDS=0.853,DPCODS=0.047,DPODUS=0.05,DTCODS=1250.,ETABCS=0.9875,
PRFDS=2.2,PRIDS=3.435,PRCDS=7.435,MAPDS=754.,MAIDS=10°.,
TFHPDS=50.0,CNHPDS=2.0,ETHPDS=0.881,TFIPDS=120.,CNIPDS=2.2,ETIPDS=.881,
TFLPDS=130.0,CNLPDS=2.3,FTLPDS=0.917,DPAFDS=0.0,CVMNOZ=0.980,
DELFG=1.0,DELFN=1.0,DELSFC=1.0,PCBLF=0.0,PCCLI=0.0,PCSLC=0.0,
T4DS=2860.,AM=0.0,ALTP=0.0\$

SLS TRISPL DESIGN POINT

\$DATAIN MODE=0,T4=2860.,ITITLE=1\$

SETUP FOR SLS AFTERBURNING

\$DATAIN T4=2860.,IAFTBN=1,T7=3700.,ETAA=0.875,ITITLE=1\$

SLS FULL AFTERBURNING

\$DATAIN MODE=3,PCNF=99.0,AM=0.75,ALTP=25000.,ITITLE=1\$

SUBSONIC POWER HOOK

\$DATAIN PCNF=90.0

\$DATAIN PCNF=80.0

\$DATAIN PCNF=70.0

\$DATAIN MODE=1,PCNC=100.,AM=1.0,ALTP=50000.,ITITLE=1\$

SETUP FOR SUPERSONIC AFTERBURNING

\$DATAIN PCNC=100.,IAFTBN=1,T7=3700.,ETAA=0.85,ITITLE=1\$

SUPERSONIC FULL AFTERBURNING

\$DATAIN IEND=1\$

SECTION IX

SAMPLE OUTPUT LISTING

The following are typical computer printouts, the first point is the design point and includes a page of correction (or scaling) factors and a page of values of variables in common. The other operating points consist of a page of output for each point. Not included for these points is a common dump, which follows each output page and is very similar to the common dump following the design point correction factors.

POINT

| | | | | | | | | |
|-------------------------------------|---------|---------------|---------|---------------|--------------|---------------|----------|---------------|
| FAN DESIGN | PRFCF= | .30001200E+01 | ETAFCF= | .95341519E+00 | MAFCF= | .58999571E+00 | T2DS= | .51866820E+03 |
| I.P. COMPRESSOR DESIGN | FRICF= | .41531922E+01 | ETAICF= | .97287007E+00 | MAICF= | .18585113E+00 | T21DS= | .67476966E+03 |
| H.P. COMPRESSOR DESIGN | PRCCF= | .34787092E+00 | ETACCF= | .99186872E+00 | MACCF= | .19867858E+00 | T22DS= | .10024490E+04 |
| COMBUSTOR DESIGN | MA3CDS= | .18833805E+02 | ETABCF= | .98750000E+00 | CTCOCF= | .89568366E+00 | | |
| H.P. TURBINE DESIGN | CMHPCF= | .10695794E+01 | TFHPCF= | .30748997E+01 | ETHPCF= | .97888889E+00 | DHHPCF= | .67288169E+00 |
| I.P. TURBINE DESIGN | CNIPCF= | .10950379E+01 | TFIPCF= | .38446973E+01 | EIIPCF= | .97888889E+00 | DHIPCFC= | .14359988E+01 |
| L.P. TURBINE DESIGN | CNLPFC= | .10820381E+01 | TFLPCF= | .25251660E+01 | ETLPCF= | .10394036E+01 | DHLPCF= | .31564692E+01 |
| DUCTY DESIGN | MA23DS= | .25046259E+04 | | | | | | |
| TURBINE/DUCT AREA DESIGN | A55= | .18480091E+01 | AM55= | .65168846E+00 | A25= | .64758021E+01 | AM25= | .25239827E+00 |
| AFTERBURNER ENTRANCE DESIGN AREA A6 | | 11.620 | | | | | | |
| AFTERBURNER DESIGN | MG6CDS= | .51899812E+04 | | | | | | |
| NOZZLE DESIGN | A8= | .46147586E+01 | AM8= | .10000080E+01 | A9= | .46825091E+01 | AM9= | .11367079E+01 |
| MAIN SHOCK OUTSIDE C-D NOZZLE | | FG= 17454.88 | | | FN= 17454.88 | | SFC= | .52412 |

DIVERGED AFTER 1 LOOPS

| | | | | | | | | | |
|------------|------------|------------|------------|------------|------------|------------|------------|------------|------------|
| 100000E+03 | 833300E+00 | 100000E+03 | 814300E+00 | 100000E+03 | 814300E+00 | 100000E+03 | 814300E+00 | 100000E+03 | 286000E+04 |
| 833300E+00 | 100000E+03 | 220000E+01 | 286000E+04 | 0. | 100000E+01 | 0. | 100000E+01 | 100000E+01 | 100000E+01 |
| 214300E+00 | 100000E+03 | 343500E+01 | 653000E+00 | 354000E+03 | 653000E+03 | 100000E+03 | 425319E+01 | 953415E+00 | 583996E+00 |
| 814300E+00 | 100000E+03 | 343500E+01 | 653000E+00 | 100000E+03 | 100000E+03 | 100000E+03 | 347871E+00 | 972070E+00 | 185851E+00 |
| 286000E+04 | 0. | 125000E+04 | 307500E+01 | 108338E+02 | 470000E-01 | 895684E+00 | 470000E-01 | 991869E+00 | 198679E+00 |
| 500000E+02 | 200000E+01 | 861000E+00 | 307900E+01 | 106958E+01 | 978889E+00 | 672802E+00 | 978889E+00 | 672802E+00 | 518666E+03 |
| 120000E+03 | 220000E+01 | 861000E+00 | 384470E+01 | 109504E+01 | 978889E+00 | 143600E+01 | 103940E+01 | 143600E+01 | 674770E+03 |
| 130000E+03 | 230000E+01 | 517000E+00 | 252517E+01 | 108204E+01 | 103940E+01 | 315647E+01 | 500000E-01 | 315647E+01 | 100245E+04 |
| 0. | 0. | 0. | 0. | 290463E+04 | 500000E-01 | 0. | 0. | 0. | 0. |
| 0. | 0. | 0. | 0. | 518998E+04 | 0. | 0. | 0. | 0. | 0. |
| 184001E+01 | 647580E+01 | 116201E+02 | 116201E+02 | 461476E+01 | 468251E+01 | 0. | 0. | 0. | 0. |
| 200000E+01 | 651698E+00 | 0. | 950000E+00 | 0. | 0. | 0. | 0. | 0. | 0. |
| 518670E+03 | 100000E+01 | 123918E+03 | 159103E+01 | 518666E+03 | 100000E+01 | 123918E+03 | 100000E+01 | 123918E+03 | 159103E+01 |
| 674770E+03 | 220000E+01 | 161385E+03 | 160010E+01 | 100245E+04 | 755700E+01 | 241496E+03 | 755700E+01 | 241496E+03 | 161215E+01 |
| 146442E+04 | 259583E+02 | 359655E+03 | 162430E+01 | 286000E+04 | 247383E+02 | 774590E+03 | 247383E+02 | 774590E+03 | 182875E+01 |
| 247750E+04 | 119953E+02 | 659137E+03 | 183511E+01 | 221324E+04 | 687363E+01 | 560868E+03 | 687363E+01 | 560868E+03 | 183992E+01 |
| 179758E+04 | 262345E+01 | 460833E+03 | 184603E+01 | 0. | 0. | 0. | 0. | 0. | 0. |
| 100000E+01 | 220000E+01 | 839000E+00 | 600004E+03 | 354000E+03 | 600004E+03 | 0. | 0. | 0. | 0. |
| 100000E+01 | 343500E+01 | 853000E+00 | 301279E+03 | 108000E+03 | 108000E+03 | 0. | 0. | 0. | 0. |
| 100000E+01 | 343500E+01 | 853000E+00 | 301279E+03 | 108000E+03 | 108000E+03 | 0. | 0. | 0. | 0. |
| 200000E+01 | 881000E+00 | 403681E-01 | 115453E+02 | 0. | 0. | 0. | 0. | 0. | 0. |
| 200000E+01 | 881000E+00 | 315920E-01 | 782690E+02 | 0. | 0. | 0. | 0. | 0. | 0. |
| 230000E+01 | 917000E+00 | 542121E-01 | 119985E+03 | 0. | 0. | 0. | 0. | 0. | 0. |
| 110541E+03 | 235298E-01 | 110541E+03 | 235298E-01 | 110541E+03 | 235298E-01 | 0. | 0. | 0. | 0. |
| 0. | 100000E+01 | 833300E+00 | 100000E+03 | 814300E+00 | 100000E+03 | 814300E+00 | 100000E+03 | 814300E+00 | 100000E+03 |
| 0. | 500000E+02 | 128000E+03 | 130000E+03 | 0. | 0. | 0. | 0. | 0. | 0. |
| 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. |
| 100000E+01 | 354000E+03 | 108000E+03 | 108000E+03 | 0. | 0. | 0. | 0. | 0. | 0. |
| 241496E+03 | 359655E+03 | 674770E+03 | 220000E+01 | 161385E+03 | 160010E+01 | 161385E+03 | 160010E+01 | 161385E+03 | 160362E+01 |
| 674770E+03 | 220000E+01 | 161385E+03 | 160010E+01 | 674770E+03 | 209000E+01 | 161385E+03 | 209000E+01 | 161385E+03 | 160362E+01 |
| 674770E+03 | 209000E+01 | 161385E+03 | 160362E+01 | 0. | 0. | 0. | 0. | 0. | 0. |
| 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. |
| 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. |
| 179758E+04 | 262345E+01 | 460833E+03 | 184603E+01 | 674770E+03 | 209000E+01 | 161385E+03 | 209000E+01 | 161385E+03 | 160362E+01 |
| 254122E+01 | 110541E+03 | 235298E-01 | 110541E+03 | 246000E+03 | 246000E+03 | 246000E+03 | 246000E+03 | 246000E+03 | 246000E+03 |
| 104745E+04 | 222336E+01 | 254240E+03 | 170844E+01 | 254240E+03 | 222336E+01 | 254240E+03 | 222336E+01 | 254240E+03 | 170844E+01 |
| 104745E+04 | 222336E+01 | 254240E+03 | 170844E+01 | 254240E+03 | 222336E+01 | 254240E+03 | 222336E+01 | 254240E+03 | 170844E+01 |
| 356541E+03 | 0. | 356541E+03 | 717858E-02 | 356541E+03 | 0. | 356541E+03 | 0. | 356541E+03 | 717858E-02 |
| 213757E+01 | 375174E+03 | 240000E+00 | 103628E+04 | 213745E+01 | 375174E+03 | 240000E+00 | 103628E+04 | 213745E+01 | 375174E+03 |
| 880181E+03 | 118643E+01 | 144567E+04 | 100000E+01 | 144567E+04 | 100000E+01 | 144567E+04 | 100000E+01 | 144567E+04 | 113671E+01 |
| 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. | 0. |
| 174549E+05 | 0. | 254122E+01 | 356541E+03 | 174549E+05 | 0. | 174549E+05 | 0. | 174549E+05 | 524116E+00 |

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SETUP FOR SLS AFTERBURNING

| OUTPUT | AME | 0.000 | ALTP= | 0. | ZF | Y4= | 2660.00 | WAFG | ETAR= | 1.0000 |
|-------------------------------|-------------|--------|-------------|--------|-------------|--------|-------------|--------|-------------|--------|
| PCNF | .100000E+03 | CNF | .833000E+00 | PRF | .220000E+01 | WAFG | .600000E+03 | MAF | .354000E+03 | |
| PCNI | .100000E+03 | CNI | .814300E+00 | PRI | .343500E+01 | MAIC | .301279E+03 | MAI | .108000E+03 | |
| PCMC | .100000E+03 | CNC | .814300E+00 | PRC | .343500E+01 | WACC | .100000E+03 | MAC | .106000E+03 | |
| Y2 | .518660E+03 | P2 | .674770E+03 | T21 | .220000E+01 | T22 | .100245E+04 | P22 | .755700E+01 | |
| Y3 | .146442E+04 | P3 | .259583E+02 | BLI | .220000E+01 | BLC | 0. | BLDU | 0. | |
| BL0B | 0. | BLH? | 0. | OLLP | 0. | T4 | 0. | P4 | 0. | |
| WA3 | .108005E+03 | WFB | .254122E+01 | FAR4 | .235298E-01 | ETAB | .266000E+04 | OPCOM | .247383E+02 | |
| YFFHP | .500000E+02 | CNHP | .403681E-01 | DHTC | .115453E+03 | T45 | .987500E+04 | P45 | .470000E-01 | |
| YFFIP | .128000E+03 | CNIP | .315200E-01 | DHTI | .782698E+02 | Y5 | .247750E+04 | P5 | .119953E+02 | |
| YFFLP | .139000E+03 | CNLP | .542121E-01 | DHTF | .119985E+03 | T55 | .221324E+04 | P55 | .667363E+01 | |
| T24 | .674770E+03 | F24 | .289000E+01 | WFO | 0. | W24 | .179750E+04 | FAR24 | .262349E+01 | |
| T25 | .674770E+03 | P25 | .209000E+01 | ETAD | 0. | AM25 | .246000E+03 | AM55 | 0. | |
| ETAF | .859000E+00 | ETAT | .853800E+00 | ETATHP | .881000E+00 | ETATIP | .252415E+00 | ETATLP | .652110E+00 | |
| T6 | .104745E+04 | P6 | .223588E+01 | AM6 | .239991E+08 | V6 | .881000E+00 | WG6 | .917000E+00 | |
| T7 | .104745E+04 | WFA | .717858E-02 | ETAA | 0. | OPAFI | .375160E+03 | PS8 | .356541E+03 | |
| AM8 | .100000E+01 | V8 | .144567E+04 | AM9 | .113624E+01 | V9 | .186643E+01 | PS28 | .118643E+01 | |
| AM28 | 0. | V28 | 0. | AM29 | .174286E+05 | V29 | .160485E+04 | BYPASS | .227788E+01 | |
| HPEXT | 0. | WFT | .254122E+01 | VA | 0. | FRD | 0. | CVMN07 | .980000E+00 | |
| VJM | .357275E+04 | CVMNOZ | 0. | FGM | .174286E+05 | FGP | .346999E+01 | A55 | .184001E+01 | |
| A25 | .647580E+01 | A6 | .116201E+02 | A9 | .468251E+01 | A28 | 0. | A29 | 0. | |
| MAIN SHOCK OUTSIDE C-D NOZZLE | | FGF= | 17432.11 | FN= | 17432.11 | SFC= | .5241 | | | |

SLS FULL AFTERBURNING

| NOZZLE DESIGN | A6= | .11176959E+02 | AM8= | .10000000E+01 | A9= | .111907...E+02 | AM9= | .10347625E+01 | | | | |
|-------------------------------|-------|---------------|-------|---------------|--------|----------------|---------|---------------|---------|-------------|------------|-------------|
| OUTPUT | AM= | 0.000 | ALTP= | 0. | ZF | T4= | 2860.00 | T7= | 3700.00 | ETAR= | 1.0000E+01 | |
| PCNF | CNF | .100000E+03 | CHX | .100000E+01 | ZI | .833300E+00 | PRF | .220000E+01 | WAF | .354000E+03 | MAI | .108000E+03 |
| PCNY | CHX | .100000E+03 | CNC | .100000E+01 | ZC | .614300E+00 | PRC | .343500E+01 | MAC | .108000E+03 | P22 | .108000E+03 |
| PCNC | CNC | .100000E+03 | F2 | .100000E+01 | F21 | .814300E+00 | PR2 | .343500E+01 | P22 | .108000E+03 | BLDU | .755700E+01 |
| T2 | F2 | .100000E+03 | F3 | .100000E+01 | BLF | .674770E+03 | 9LI | .220000E+01 | 8LC | .100245E+04 | P4 | 0. |
| T3 | F3 | .100000E+03 | BLMP | .259543E+02 | 9LIP | 0. | 8LLP | 0. | T4 | 0. | DPCON | .247363E+02 |
| MA3 | MA3 | .168000E+03 | MA4 | .110541E+03 | MA4 | .110541E+03 | FAR4 | .235290E-01 | ETAB | .286000E+04 | P45 | .470000E-01 |
| TFAP | CHXP | .500000E+02 | CNIP | .200000E+01 | DHTC | .403861E-01 | DHTC | .115453E+03 | T45 | .247750E+04 | P5 | .119953E+02 |
| TFIP | CNIP | .120000E+03 | CHLP | .200000E+01 | DHTCIP | .315920E-01 | DHTI | .702600E+02 | T5 | .221324E+04 | P55 | .687363E+01 |
| TFEL | CHLP | .130000E+03 | F24 | .230000E+01 | DHTCLP | .542321E-01 | DHTF | .119905E+03 | T55 | .179750E+04 | FAR24 | .262349E+01 |
| T24 | F24 | .874770E+03 | P25 | .209000E+01 | WAD | .246600E+03 | MFD | 0. | AM25 | .246000E+03 | AM55 | 0. |
| T25 | P25 | .674770E+03 | ETAI | .209000E+01 | DPOVC | .500000E-01 | ETAN | 0. | ETATIP | .252415E+00 | ETAILP | .652110E+00 |
| ETAF | ETAI | .839000E+00 | P6 | .253000E+00 | PS6 | .053000E+00 | ETATHP | .261000E+00 | V6 | .661000E+00 | MG5 | .917000E+00 |
| T6 | P6 | .104745E+04 | PSA | .183777E+02 | FAR7 | .213765E+01 | AM6 | .239991E+00 | OPAFI | .375160E+03 | PSR | .356541E+03 |
| T7 | PSA | .370000E+04 | VB | .256563E+04 | PS9 | .605055E-01 | ETAA | .875000E+00 | V9 | 0. | PS28 | .104245E+01 |
| AM8 | VB | .108000E+01 | V23 | .116201E+02 | PS29 | .100000E+11 | AM9 | .103476E+01 | V29 | 0. | BYPASS | .22778E+01 |
| AM28 | V23 | 0. | MFT | .214189E+02 | MGY | 0. | VA | 0. | FRD | 0. | CVMNOZ | .908000E+00 |
| WJM | CYNOZ | .269251E+04 | AM | .647536E+01 | VJD | .375419E+03 | FGM | .314172E+05 | FGP | 0. | A55 | .104001E+01 |
| A25 | AM | 0. | A6 | .116201E+02 | AM | .211740E+02 | AS | .111907E+02 | A28 | 0. | A29 | 0. |
| MAIN SHOCK OUTSIDE C-D NOZZLE | SFC= | 2.45433 | FM= | 31417.20 | FN= | 31417.20 | | | | | | |

OUTPUT

| | | | |
|-------------------------------|---------------|-------------|--------------|
| ALM= .750 | ALTIV= 25000. | TA= 2025.19 | ETAR= 1.0000 |
| PCNF | CNF | PRF | MAF |
| .990000E+02 | .103121E+01 | .233598E+01 | .206372E+03 |
| PCNI | CNI | PRI | MAI |
| .597000E+02 | .102371E+01 | .356003E+01 | .657156E+02 |
| PCNC | CNC | PRC | MAC |
| .980020E+02 | .100375E+01 | .340908E+01 | .656722E+02 |
| T2 | P2 | P21 | P22 |
| .478044E+03 | .539056E+00 | .641051E+03 | .449396E+01 |
| T3 | P3 | BLI | BLDU |
| .142966E+04 | .156834E+02 | 0. | 0. |
| BLOR | SLMP | BLIP | T4 |
| 0. | 0. | 0. | .149504E+02 |
| WA3 | WFB | MG4 | DPCOM |
| .656722E+02 | .153855E+01 | .672107E+02 | .467386E-01 |
| TFFHP | CMHP | DHTCHP | P45 |
| .500210E+02 | .198018E+01 | .403839E-01 | .724009E+01 |
| TFFIP | CNIP | DHTCIP | P5 |
| .120166E+03 | .220905E+01 | .321475E-01 | .409820E+01 |
| TFFLP | CNLP | DHTCLP | P55 |
| .131393E+03 | .229425E+01 | .550116E-01 | .154208E+01 |
| T24 | P24 | MA0 | FAR24 |
| .641051E+03 | .119792E+01 | .140656E+03 | 0. |
| T25 | P25 | DPDUC | AMS5 |
| .641051E+03 | .119792E+01 | .486233E-01 | .682441E+00 |
| ETAF | ETAI | ETAC | ETATLP |
| .305772E+00 | .836766E+00 | .849513E+00 | .919003E+00 |
| T6 | P6 | PS6 | MG6 |
| .102871E+04 | .128425E+01 | .123459E+01 | .207867E+03 |
| T7 | WFA | FAR7 | PS8 |
| .102871E+04 | 0. | .745684E-02 | .685188E+00 |
| AM8 | V8 | PS9 | PS28 |
| .100000E+01 | .143283E+04 | .577783E+00 | 0. |
| AM28 | V28 | PS29 | BYPASS |
| 0. | 0. | 0. | .214037E+01 |
| WEXT | NFT | MG7 | CVMNOZ |
| 0. | .153855E+01 | .207510E+03 | .980000E+00 |
| WJM | CVMNOZ | VJD | A55 |
| .155875E+04 | 0. | 0. | .184001E+01 |
| A25 | A6 | A8 | A29 |
| .647508E+01 | .116201E+02 | .461476E+01 | 0. |
| MAIN SHOCK OUTSIDE C-0 NOZZLE | FG= 12118.73 | FN= 7228.39 | SFC= .76626 |

SUBSONIC POWER NOOK

| OUTPUT | APR .750 | ALTP= 25000. | T4= 2505.19 | ETAR= 1.0000 |
|-------------------------------|-------------|--------------|-------------|--------------|
| PCNF | .937461E+00 | ZF | PRF | MAF |
| PCNI | .991261E+00 | ZI | FRI | .106295E+03 |
| PCNC | .932756E+02 | ZC | FRC | .540067E+02 |
| T2 | .470944E+03 | T21 | F21 | .539977E+02 |
| T3 | .129033E+04 | BLF | BLI | .360276E+01 |
| BL02 | 0. | BLIF | BLP | 0. |
| MA3 | .539977E+02 | MG4 | FAR4 | .115277E+02 |
| IFHP | .500143E+02 | DHTCHP | DHTC | .474593E-01 |
| IFFI | .129217E+03 | DHTCIP | DHTI | .555067E+01 |
| IFFL | .129702E+03 | DHTCLP | DHTF | .320299E+01 |
| T24 | .643440E+03 | MAD | MFD | .125485E+01 |
| T25 | .063440E+03 | OPDUC | OPDU | .132280E+03 |
| ETAF | .843004E+00 | ETAC | ETATHP | .261722E+00 |
| T6 | .9.1678E+03 | PS6 | AMG | .915050E+00 |
| T7 | .9616.3E+03 | FAR7 | ETA | .107344E+03 |
| AM0 | .100000E+01 | PS9 | CPAFI | .575798E+00 |
| AM7 | 0. | PS29 | V9 | .169032E+04 |
| HPEXT | 0. | MG1 | V29 | 0. |
| VJM | .146851E+04 | VJD | FRO | .244948E+01 |
| AZ5 | .647558E+01 | A8 | CVMNOZ | .900000E+00 |
| MAIN SHOCK OUTSIDE C-D NOZZLE | | PS29 | FGP | .A55 |
| | | A9 | A28 | .184001E+01 |
| | | A28 | A29 | 0. |
| | | | | SFC= .72926 |
| | | | | FN= 5226.20 |
| | | | | PU= 9640.79 |

SUBSONIC POWER HOOK

| OUTPUT | ALTP= 25000. | ZF | T4= 2259.62 | WAF | ETAR= 1.0000 |
|-------------------------------|--------------|-------------|-------------|--------|--------------|
| AM | .750 | | | | |
| PCNF | .833299E+00 | .680238E+00 | .172439E+01 | WAF | .166047E+03 |
| PCNI | .902995E+00 | .229222E+00 | .308410E+01 | WAI | .436001E+02 |
| PCNC | .976350E+00 | .780779E+00 | .322059E+01 | WAC | .436025E+02 |
| P2 | .539056E+00 | .574658E+03 | .929545E+00 | P22 | .285801E+01 |
| P3 | .926149E+01 | .0 | .0 | BLD | .0 |
| BLPB | .0 | .0 | .0 | P4 | .0 |
| WFB | .733083E+00 | .443355E+02 | .160129E-01 | ETAB | .861625E+01 |
| CMHP | .198921E+01 | .05279E-01 | .916093E+02 | T45 | .480744E-01 |
| CMIP | .207466E+01 | .302336E-01 | .580174E+02 | T5 | .424224E+01 |
| CMPL | .207995E+01 | .500575E-01 | .660027E+02 | P55 | .2480012E+01 |
| P24 | .879009E+00 | .122446E+03 | .0 | FAR24 | .101980E+01 |
| P25 | .879009E+00 | .543671E-01 | .0 | AM55 | .0 |
| ETAI | .875427E+00 | .839205E+00 | .879697E+00 | ETATLP | .556009E+00 |
| P6 | .912776E+00 | .877458E+00 | .238976E+00 | W6 | .916426E+00 |
| WFA | .0 | .441486E-02 | .0 | PS8 | .166782E+03 |
| W8 | .127256E+04 | .409222E+00 | .113661E+01 | PS28 | .484374E+00 |
| AM28 | .0 | .0 | .0 | BYPASS | .0 |
| MPEXT | .733083E+00 | .166780E+03 | .762422E+03 | CVMNOZ | .200840E+01 |
| WJM | .138415E+04 | .0 | .717506E+04 | A55 | .980000E+00 |
| A25 | .647580E+01 | .116201E+02 | .468251E+01 | A29 | .184001E+01 |
| MAIN SHOCK OUTSIDE C-D NOZZLE | FG= 7549.92 | FM= 3615.14 | SFC= .73001 | | |

SUBSONIC POWER HOOK

| OUTPUT | 750 | 25000 | 1999.81 | 1.0300 |
|-------------|-------------|-------------|-------------|-------------|
| PCNF | GNF | ZF | PRF | MAF |
| .700600E+02 | .729137E+00 | .534079E+00 | .155839E+01 | .146345E+03 |
| PCNI | GNI | ZI | PRI | HAI |
| .725324E+02 | .506767E+00 | .608100E+00 | .263448E+01 | .330381E+02 |
| PCNC | CNC | ZC | PRC | HAC |
| .630559E+02 | .562114E+00 | .759621E+00 | .31513E+01 | .330334E+02 |
| T2 | F2 | T21 | P21 | P22 |
| .478044E+03 | .539056E+00 | .545438E+03 | .786153E+00 | .211827E+01 |
| T3 | P3 | BLF | BLI | BLDU |
| .308124E+04 | .658600E+01 | 0. | 0. | 0. |
| BL08 | BLHP | BLIP | BLP | P4 |
| 0. | 0. | 0. | 0. | .626534E+01 |
| WA3 | HFB | WG4 | FAR4 | DPCOM |
| .330334E+02 | .465903E+00 | .334993E+02 | .141040E-01 | .486867E-01 |
| TFHP | CNHP | DHTCHP | DHTC | P45 |
| .500338E+02 | .198650E+01 | .405510E-01 | .811664E+02 | .300474E+01 |
| TFIP | CNIP | DHTCIP | DHTI | P5 |
| .123625E+03 | .192054E+01 | .279649E-01 | .478546E+02 | .182993E+01 |
| TFPLP | CNPLP | DHTCLP | DHTF | P55 |
| .123273E+03 | .193290E+01 | .455584E-01 | .704557E+02 | .812212E+00 |
| T24 | P24 | HAD | MFD | FA.24 |
| .543436E+03 | .740601E+00 | .113307E+03 | 0. | 0. |
| T25 | C25 | OPUUC | ETAO | AM55 |
| .543436E+03 | .740601E+00 | .579652E-01 | 0. | .475891E+00 |
| ETAF | ETAI | ETAC | ETATHP | ETATLP |
| .887344E+00 | .82345E+00 | .831202E+00 | .879504E+00 | .915912E+00 |
| T6 | P6 | PS6 | AM6 | MG6 |
| .717923E+03 | .756935E+00 | .727653E+00 | .238473E+00 | .146806E+03 |
| T7 | M7 | FAR7 | ETAA | PS8 |
| .717923E+02 | 0. | .31839E-02 | 0. | .400881E+00 |
| AM9 | V9 | PC9 | AM9 | PS28 |
| .109000E+01 | .119933E+04 | .332145E+00 | .133091E+04 | 0. |
| AM28 | V28 | FSR9 | AM29 | BYPASS |
| 0. | 0. | 0. | 0. | .342959E+01 |
| HPENT | MFT | MG1 | VA | CVMNOZ |
| 0. | .465903E+00 | .146801E+03 | .762422E+03 | .980000E+00 |
| WJN | CVMNOZ | VJO | FGM | A55 |
| .139429E+04 | 0. | 0. | .595135E+04 | .184801E+01 |
| A25 | A6 | A8 | A9 | A29 |
| .647580E+01 | .116201E+02 | .462476E+01 | .468251E+01 | 0. |
| NOZZLE | FG= | FH= | FN= | SFU= |
| | 5424.85 | 2156.94 | | .77761 |

ALIP= 50000. T4= 2850.40

| ALIP= 50000. | ZF | T4= 2850.40 | PRF | ETAR= 0.445 |
|-------------------------------|--------------|-------------|--------------|-------------|
| PCNF | .795330E+00 | .545432E+00 | .63700E+01 | MAF |
| PCNI | .872474E+00 | .622443E+00 | .295360E+01 | .157100E+03 |
| PCNC | .970105E+00 | .772839E+00 | .318068E+01 | MAI |
| TE | .643321E+03 | .760192E+03 | .101658E+01 | .297357E+02 |
| T3 | .152572E+04 | .955024E+01 | .0. BLF | MAC |
| BL0E | .0. | .0. BLJF | .0. BLKJ | .297237E+02 |
| MA3 | .397237E+02 | .0. M04 | .0. FAR4 | .300258E+01 |
| TFHP | .508090E+02 | .406199E+02 | .225608E-01 | RLDU |
| TFIP | .120519E+03 | .203108E+01 | .0. WED | .0. P4 |
| TFFLP | .126723E+03 | .960806E+00 | .0. ETAD | .286440E+04 |
| T24 | .760192E+03 | .547541E-01 | .0. ETATP | .209220E+01 |
| T25 | .760192E+03 | .0. CTAX | .0. AM29 | .989220E+01 |
| ETAF | .827305E+00 | .0. PS6 | .0. AM29 | .474613E-01 |
| T6 | .106105E+04 | .0. MFT | .0. VA | .436759E+01 |
| YF | .106105E+04 | .296199E+08 | .174329E+04 | .259858E+01 |
| AM3 | .106000E+01 | .0. CVDNOZ | .0. FCM | .259858E+01 |
| AM28 | .0. | .0. A6 | .0. A9 | .109695E+01 |
| MPEXT | .0. | .116201E+32 | .461476E+01 | .109695E+01 |
| VJM | .150299E+04 | .0. A6 | .0. A9 | .109695E+01 |
| AZ5 | .647530E+01 | .0. A6 | .0. A9 | .109695E+01 |
| MAIN SHOCK OUTSIDE C-D NOZZLE | FG= 11057.94 | FN= 2545.80 | SFC= 1.26731 | |

SUPERSONIC FULL AFTERBURNING

| NOZZLE DESIGN | A8= | .11057560E+02 | AM8= | .10000000E+01 | A9= | .19339090E+02 | AM9= | .19724375E+01 |
|-------------------------------|--------|---------------|--------|---------------|--------|---------------|--------|---------------|
| OUT PUT | AM | 1.640 | ALPA | 50000. | T4= | 2060.40 | T7= | 3700.00 |
| PCNF | CNF | .685761E+02 | CNI | .795330E+00 | ZF | .645492E+00 | MAFC | .157100E+03 |
| PCNI | CNI | .926056E+02 | CNC | .672474E+00 | ZI | .163708E+01 | MAIC | .397357E+02 |
| PCMC | CNC | .180000E+03 | P2 | .978105E+00 | ZC | .822463E+00 | WACC | .397237E+02 |
| T2 | P2 | .643321E+03 | P3 | .820975E+00 | T21 | .772839E+00 | T22 | .300258E+01 |
| T3 | P3 | .152572E+04 | P4 | .955024E+01 | T22 | .760192E+03 | P22 | .300258E+01 |
| BL0B | BLHP | 0. | BLIP | 0. | BLI | .101658E+01 | BLC | 0. |
| MA3 | WFB | .397237E+02 | MG4 | .406199E+02 | 9LLP | 0. | T4 | 0. |
| TFFRP | CNHP | .500093E+02 | DHTCHP | .405684E-01 | FAR4 | .225608E-01 | ETAB | .909220E+01 |
| TFPIP | CNIP | .120519E+03 | DHTCIP | .296827E-01 | DHTC | .116074E+03 | T4E | .479613E-01 |
| TFFLP | CNLP | .126723E+03 | DHTCLP | .489704E-01 | DHTI | .734843E+02 | T5 | .438675E+01 |
| T24 | P24 | .760192E+03 | WAD | .117364E+03 | DHTF | .109147E+03 | T55 | .259658E+01 |
| T25 | P25 | .760192E+03 | DPCUC | .547941E-01 | WFD | 0. | MG24 | .109695E+01 |
| ETAF | ETAI | .827105E+00 | ETAC | .035916E+00 | ETAD | 0. | AM25 | 0. |
| T6 | P6 | .206105E+04 | PS6 | .114456E+00 | ETATHP | .800399E+00 | AM5 | .540960E+00 |
| T7 | P7 | .370000E+04 | FAR7 | .954128E+00 | AM6 | .239704E+00 | ETATLP | .915349E+00 |
| WFA | V9 | .446885E+04 | PSS | .602581E-01 | ETAA | .658000E+00 | MG6 | .157984E+03 |
| AM20 | V20 | .647588E+01 | PS29 | .114456E+00 | AM9 | .197244E+01 | PS0 | .467535E+00 |
| WPEYT | WFT | 0. | MG7 | 0. | AM29 | .455923E+04 | PS28 | 0. |
| VJM | CVDMOZ | .446885E+04 | VJD | .166571E+03 | VA | 0. | BYPASS | 0. |
| A25 | A6 | .647588E+01 | A8 | .110576E+02 | FRD | .295362E+01 | CYMMOZ | .295362E+01 |
| MAIN SHOCK OUTSIDE C-D NOZZLE | FG= | 23139.10 | FN= | 14610.84 | FGP | .851214E+04 | A55 | .980000E+00 |
| | | | | | A9 | .231302E+05 | A29 | .184801E+01 |
| | | | | | A28 | .193339E+02 | A29 | .184801E+01 |
| | | | | | SFO= | 2.33239 | | |

REFERENCES

1. John S. McKinney, Simulation of Turbofan Engine, Parts I & II, AFAPL-TR-67-125, Air Force Aero Propulsion Laboratory, Research and Technology Division, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio, November 1967
2. Laurence H. Fishbach and Robert W. Koenig, Geneng II - A Program for Calculating Design and Off-Design Performance of Two- and Three-Spool Turbofans with as Many as Three Nozzles, NASA TN D-6553, NASA-Lewis Research Center, Cleveland, Ohio, February 1972