THIS REPORT HAS BEEN DELIMITED AND CLEARED FOR PUBLIC RELEASE UNDER DOD DIRECTIVE 5200.20 AND NO RESTRICTIONS ARE IMPOSED UPON ITS USE AND DISCLOSURE.

DISTRIBUTION STATEMENT A APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED,

# SECURITY MARKING

The classified or limited status of this report applies to each page, unless otherwise marked. Separate page printouts MUST be marked accordingly.

THIS DOCUMENT CONTAINS INFORMATION AFFECTING THE NATIONAL DEFENSE OF THE UNITED STATES WITHIN THE MEANING OF THE ESPIONAGE LAWS, TITLE 18, U.S.C., SECTIONS 793 AND 794. THE TRANSMISSION OR THE REVELATION OF ITS CONTENTS IN ANY MANNER TO AN UNAUTHORIZED PERSON IS PROHIBITED BY LAW.

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.

Contract No. Nonr 839(38)



### OSCILLATIONS IN THE PLASMA SHEATH

by

Simon H. Schwartz





JULY 1965

## POLYTECHNIC INSTITUTE OF BROOKLYN

DEPARTMENT of AEROSPACE ENGINEERING and APPLIED MECHANICS

PIBAL REPORT NO. 880

#### OSCILLATIONS IN THE PLASMA SHEATH

by

#### Simon H. Schwartz

This research has been conducted in part under Contract No. Nonr 839(38) for PROJECT DEFENDER, and was made possible by the support of the Advanced Research Projects Agency under Order No. 529 through the Office of Naval Research.

Reproduction in whole or in part is permitted for any purpose of the United States Government.

#### Polytechnic Institute of Brooklyn

#### Department

of

Aerospace Engineering and Applied Mechanics

July 1965

PIBAL Report No. 880

#### OSCILLATIONS IN THE PLASMA SHEATH<sup>T</sup>

by

Simon H. Schwartz<sup>‡</sup> Polytechnic Institute of Brooklyn

#### SUMMARY

The behavior of the ion plasma sheath next to a wall is investigated under a sinusoidal-in-time perturbation of the potential of the wall. A collisionless sheath is assumed, and the ion density distribution is considered to remain at the steady state value, with only the electron distribution being perturbed. The collisionless macro-scopic plasma equations are solved by a perturbation procedure, using an exact equation for the steady state potential distribution. A second order differential equation which is linear, but which has variable coefficients is obtained for the perturbation. This equation is transformed into the form of a wave equation with variable propagation constant so that the analytical behavior can be deduced. One observes that above a certain range of frequencies, but below the plasma frequency of the uniform plasma, the perturbation may propagate over a finite region which may begin away from the wall.

A numerical integration of the equation is performed, using an asymptotic approximation in order to obtain the boundary conditions. The predicted oscillations

<sup>‡</sup>Research Associate

<sup>&</sup>lt;sup>†</sup>This research has been conducted in part under Contract No. Nonr 839(38) for PROJECT DEFENDER, and was made possible by the support of the Advanced Research Projects Agency under Order No. 529 through the Office of Naval Research.

are seen to occur, and two resonances in the sheath are found at frequencies below the plasma frequency.

s,

### TABLE OF CONTENTS

Section		Page
I	Introduction	1
ш	Basic Equations	3
ш	Derivation of Perturbation Equation for E	6
IV	Transformation of the Perturbation Equation	8
v	Results	10
VI	Conclusions	12
VII	References	12
	Appendix	14

## LIST OF ILLUSTRATIONS

Figure	<u>1</u>	Page
1	Potential Distribution as a Function of Distance From Wall	15
2	Propagation Factor ¥ vs. Distance From Wall	
3	Electric Field Distribution as a Function of Distance	-0
	From Wall	17

#### SECTION I

#### INTRODUCTION

The physics of plasma adjacent to a boundary, the so-called "plasma sheath," is not completely understood. Yet boundary phenomena play an essential role in numerous well-known situations, for example, those associated with Langmuir Probes, resonance probes and communication antennae.

A few of the numerous papers dealing with the steady-state behavior of the sheath are listed in the bibliography. With respect to time-dependent behavior, experimental studies of natural oscillations in the sheath have been performed by Gaby - Ash, and Dracott<sup>1</sup>, by Ott, Gierke and Schwirzke<sup>2</sup>, and by Harp and Kino<sup>3</sup> with combicting results. The resonance probe employing appliedoscillations has been studied by Wimmel<sup>4</sup>, and the theory of the RF plasma sheath has been studied by Pavkovich and Kino<sup>5</sup> and Pavkovich<sup>6</sup>. The latter approach the problem theoretically by integrating the collisionless Boltzmann equation numerically, assuming a parabolic steady-state potential distribution, a harmonic time-wise perturbation in the plasma and in the consistent boundary conditions, and a time-independent perturbation amplitude. They reach the conclusion that no spatial oscillations about zero electric field are possible in the sheath when the electron density distribution is Maxwellian. The behavior of temporally amplified or damped perturbations is not included in their studies. Pavkovich<sup>6</sup> also investigates a theory similar to the one to be set forth in this paper, i.e., a fluid dynamical approach, but with a parabolic potential distribution. However, he discounts it because of the possibility of waves (spatial) near the walls. Moreover,

on the basis on an examination of the so-called "sheath impedance", he decides that no oscillations are possible, at least in the numerical cases considered.

In the present paper, an ion plasma sheath next to a solid boundary is considered. This boundary can be considered to be insulated with respect to ground; such a boundary is termed "floating". It is well-known that the plasma region adjacent to a floating wall is characterized by a large potential drop. In particular, a large negative potential and negative surface charge is created, since the electrons of the plasma are more mobile than the ions and diffuse to the wall more readily. The' resulting preponderance of ions and the associated positive space charge in this neighborhood suggest the designation "ion sheath".

The plasma sheath is assumed to be subjected to a temporally oscillatory perturbation of the surface potential. The amplitude of this oscillation is assumed to be time-wise invariant both at the boundary and in the plasma. In another view we may consider the oscillation to have appeared in the plasma, whereas the surface oscillation follows as a result. Finally, this type of problem may be interpreted as an investigation of the existence and spatial properties of a neutral disturbance (undamped and unamplified) consistent with the boundary conditions.

In this paper, it is assumed that collisions between charged particles within this sheath are negligible. The collisionless nonlinear macroscopic plasma equations are taken to be applicable here. The equations are linearized by a perturbation method and an equation for the perturbation derived. It is assumed that since the electrons are so much more mobile than the ions, only the electron density distribution will be perturbed. Under the conditions of this problem the electron energy equation can be

integrated to give an adiabatic equation of state for the electron gas. However, the value of the adiabatic exponent  $\gamma$  is not specified at the outset, although it is clear that if the frequency is high enough, there will be insufficient time for equipartition of energy between the  $\alpha$ -grees of freedom so that  $\gamma = 3$ , while for low frequencies  $\gamma = \frac{5}{3}$  is the correct value.

In this particular analysis a plane wall is treated and spatial variations in only one direction are considered, so that the problem is one-dimensional.

#### SECTION II

#### BASIC EQUATIONS

The basic equations of the problems are:

$$n_{e} \frac{\partial \underline{u}}{\partial t} = \frac{-q}{m} n_{e} \underline{E}^{*} - \frac{1}{m} \nabla p_{e}$$
(1)

$$\nabla \cdot \underline{\underline{E}}^{*} = \frac{-\mathbf{q}}{\epsilon_{o}} (\mathbf{n}_{e} - \mathbf{n}_{i})$$
(2)

$$p_e n_e^{-\gamma} = const.$$
 (3)

$$\nabla x \underline{H} = \varepsilon_0 \frac{\partial \underline{E}}{\partial t} - n_e q \underline{u}^*$$
(4)

$$\nabla x \overline{\underline{E}} = -u_0 \frac{\partial \overline{\underline{H}}}{\partial t}$$
(5)

where  $\underline{u}^*$  is the electron velocity,  $\underline{E}^*$  is the total electric field,  $\underline{p}_e$  is the total electron pressure,  $\underline{n}_e$  and  $\underline{n}_i$  are the total electron and ion number densities respectively,  $\underline{\overline{E}}$ and  $\underline{\overline{H}}$  are the perturbation electric and magnetic fields,  $\gamma$  is the adiabatic exponent,

and n is the steady state electron number density. Equation (3) is used since one is concerned with high frequency oscillations and because in this case the energy equation can be solved to obtain the adiabatic equation of state. In Eq. (1) the nonlinear acceleration term  $\underline{u}^* \cdot \nabla \underline{u}^*$  is left out since it is assumed to be small compared with the time derivative. In this set of equations  $\underline{u}^*$  is a perturbation already, since the steady state velocity is zero for an ion sheath (the electrons are reflected back in the negative potential so that their average velocity is zero).

Now, let  $\underline{E}^*$  and  $\underline{n}_e$  be perturbed so that  $\underline{E}^* = \underline{E}_0 + \underline{\overline{E}}$  and  $\underline{n}_e = \underline{n}_e + \overline{n}$  where  $|\overline{\underline{E}}_0| <<1$  and  $|\overline{\underline{n}}_n| <<1$  where  $\underline{E}_0$  and  $\underline{n}_e$  are the steady state solutions and  $\underline{\overline{E}}$  and  $\overline{\overline{n}}$  are the perturbations. If this perturbation is substituted in Poisson's equation, Eq. (2), it becomes

$$\nabla \cdot (\underline{\mathbf{E}}_{o} + \overline{\underline{\mathbf{E}}}) = \frac{-\mathbf{q}}{\varepsilon_{o}} (\mathbf{n}_{e} + \overline{\mathbf{n}} \cdot \mathbf{n}_{i}).$$
(6)

The steady state Poisson's equation is

$$\nabla \cdot \underline{\mathbf{E}}_{o} = \frac{-\mathbf{q}}{\epsilon_{o}} (\mathbf{n}_{e_{o}} - \mathbf{n}_{i})$$
(7)

In the one-dimensional case this equation becomes (see Appendix)

$$\frac{\mathrm{d}\theta}{\mathrm{d}\xi}^{2} = 2\sqrt{1-2\theta} + 2e^{\theta} - 4$$
(8)

where  $\theta = \frac{q\varphi}{kT_o}$ ,  $\varepsilon = \frac{x}{\lambda_d}$ , where  $E_o = -\frac{d\varphi}{dx}$  and  $\lambda_d$  is the Debye length referred to undisturbed plasma,  $\lambda_d^2 = \frac{\varepsilon_o k T_o}{n_o q^2}$ . Here,  $T_o$  is the electron temperature of the undisturbed plasma and k is the Boltzmann constant. Thus, Eq. (6) becomes

$$\frac{dE}{dx} = \frac{-q}{\varepsilon_0} \frac{\pi}{n}$$
(9)

Let the adiabatic equation of state Eq. (3) be examined more closely.

$$p_{e}n_{e}^{-\gamma} = p_{e}n_{e}^{-\gamma} = kT_{o}n_{e}^{1-\gamma}$$
(10)

where  $p_{o} = n_{o} kT_{o}$  is the isothermal equation of state and holds for steady state,

$$p_e = kT_0 n_e^{1-\gamma_n \gamma}$$
(10a)

Now, let one substitute the perturbation into (10a). Then (10a) becomes

$$p_{e} = kT_{o}n_{e_{o}}^{1-\gamma}n_{e_{o}}^{\gamma}(1+\frac{\bar{n}}{n_{e_{o}}})^{\gamma} = kT_{o}n_{e_{o}}(1+\frac{\gamma\bar{n}}{n_{e_{o}}}) = kT_{o}n_{e_{o}}^{\gamma} + \gamma kT_{o}\bar{n}$$
(11)

$$\nabla \mathbf{p}_{e} = \mathbf{k} \mathbf{T}_{o} \nabla \mathbf{n}_{e} + \gamma \mathbf{k} \mathbf{T}_{o} \nabla \overline{\mathbf{n}}$$
(12)

$$\frac{1}{n_e} \nabla p_e = \frac{1}{(n_e + \overline{n})} \left( kT_o \nabla n_e + \gamma kT_o \nabla \overline{n} \right) = \frac{\gamma kT_o}{n_e} \nabla \overline{n} + \frac{kT_o}{n_e} \left( 1 - \frac{\overline{n}}{n_e} \right) \nabla n_e$$
(13)

With Eq. (13), Eq. (1) becomes

$$m \frac{\partial \underline{u}}{\partial t} = -q \left( \underline{E}_{o} + \underline{\overline{E}} \right) - \frac{\gamma k T_{o}}{n_{e_{o}}} \nabla \overline{n} - \frac{k T_{o}}{n_{e_{o}}} \left( 1 - \frac{\overline{n}}{n_{e_{o}}} \right) \nabla n_{e_{o}}$$
(14)

$$n_{e_{o}} \frac{\lambda \underline{u}}{\partial t} = \frac{-qn_{e}}{m} \underbrace{E}_{o} - \frac{qn_{e}}{m} \underbrace{E}_{o} \frac{\gamma kT}{m} \sqrt{n} - \frac{kT}{m} \sqrt{n} e_{o} + \frac{kT}{m} \frac{\sqrt{n}}{n} e_{o} \frac{kT}{n}$$
(15)

The steady state momentum equation is

$$\frac{-qn_e}{m} = \frac{E}{m} - \frac{kT_o}{m} \nabla n_e = 0$$
(16)

which has a solution

$$n_{e} = n_{o}^{e}$$
(17)

where  $\underline{\mathbf{E}}_{\mathbf{0}} = -\nabla \varphi$  and  $\mathbf{n}_{\mathbf{0}}$  is the value of the electron density at  $\varphi = 0$ . Thus, Eq. (15) becomes

$$n_{e_{o}}\frac{\partial u}{\partial t} = -\frac{q}{m} n_{e_{o}} \overline{E} - \frac{\gamma kT_{o}}{m} \nabla \overline{n} + \frac{kT_{o}}{m} \frac{\nabla n_{e}}{n_{e_{o}}} \overline{n}$$
(18)

which with the  $e^{j\omega t}$  time dependence becomes

$$jwn_{e_{o}} \underline{u} = \frac{-q}{m}n_{e_{o}} \underline{E} - \frac{\gamma kT_{o}}{m} \nabla n + \frac{kT_{o}}{m} \frac{\nabla n}{n_{e_{o}}} n \qquad (18a)$$

#### SECTION III

## DERIVATION OF PERTURBATION EQUATION FOR E

Let one substitute Eq. (14a) into Maxwell Eq. (4).

$$\nabla \mathbf{x} \underline{\mathbf{H}} = j \omega \varepsilon_{0} \underline{\mathbf{E}} + \frac{\mathbf{q}^{2}}{j \omega m} \mathbf{n}_{0} \underline{\mathbf{E}} + \frac{\mathbf{q}}{j \omega} \frac{\gamma \mathbf{k} \mathbf{T}_{0}}{m} \nabla \mathbf{n} - \frac{\mathbf{q}}{j \omega} \frac{\mathbf{k} \mathbf{T}_{0}}{m} \frac{\nabla \mathbf{n}_{0}}{\mathbf{n}_{0}} \mathbf{n} \qquad (19)$$

$$= jw\varepsilon_{o} \frac{E}{E} + \varepsilon_{o} \frac{w}{jw} \frac{E}{jw} + \frac{a^{2}q}{jw} \nabla n - \frac{a^{2}q}{\gamma jw} \frac{e_{o}}{n} n \qquad (19a)$$

where 
$$a^{2} = \frac{\gamma k T_{o}}{m}$$
 and  $w_{p}^{2} = \frac{n_{e} q^{2}}{\epsilon_{o} m}$ .

$$jw\nabla x \underline{H} = -\varepsilon_{o}(w^{2} - w_{p}^{2}) \underline{E} + a^{2}q \nabla n - \frac{a^{2}q}{\gamma} \left(\frac{\nabla n}{n} - \frac{\sigma}{n}\right) n \qquad (19b)$$

Making use of Eq. (18), this becomes

$$j\omega\nabla x \underline{H} = -\varepsilon_{o}(\omega^{2} - \omega_{p}^{2}) \underline{E} - \varepsilon_{o}a^{2}\nabla(\nabla \cdot \underline{E}) + \frac{\varepsilon_{o}a^{2}}{\gamma}(\frac{\nabla n}{e_{o}})\nabla \cdot \underline{E}$$
(19c)

Now, we can make use of the vector identity

$$\nabla (\nabla \cdot \underline{E}) = \nabla x (\nabla x \underline{E}) + \nabla \cdot (\nabla \underline{E})$$
(19d)

$$j\omega\nabla x \underline{H} = -\varepsilon_{o}(\omega^{2} - \omega_{p}^{2}) \underline{E} - \varepsilon_{o}a^{2}\nabla x(\nabla x \underline{E}) - \varepsilon_{o}a^{2}\nabla . (\nabla \underline{E}) + \frac{\varepsilon_{o}a^{2}}{\gamma} (\frac{\nabla n}{n_{e}})\nabla \cdot \underline{E}$$
(19e)

which by Eq. (5) becomes

$$j\omega\nabla x \underline{H} = -\varepsilon_{o}(\omega^{2} - \omega_{p}^{2}) \underline{E} + \varepsilon_{o}a^{2}j\omega_{u_{o}}\nabla x \underline{H} - \varepsilon_{o}a^{2}\nabla \cdot (\nabla \underline{E}) + \frac{\varepsilon_{o}a^{2}}{\gamma} (\frac{\nabla n}{e_{o}}) \nabla \cdot \underline{E}$$
(19f)

If one assumes that H = 0, this becomes

$$-\varepsilon_{o}(w^{2}-w^{2})\underline{E}-\varepsilon_{o}a^{2}\nabla\cdot(\nabla\underline{E})+\frac{\varepsilon_{o}a^{2}}{\gamma}(\frac{\nabla n}{n_{e}})\nabla\cdot\underline{E}=0$$
 (20)

which can be written in the form

$$\nabla \cdot (\nabla \underline{E}) - \frac{1}{\gamma} \frac{\nabla n}{n_e} \nabla \cdot \underline{E} + (\frac{\omega^2 - \omega^2}{a^2}) \underline{E} = 0$$
 (29a)

In the one-dimensional case this becomes

$$\frac{d^{2}E}{dx^{2}} - \frac{1}{\gamma} \frac{\nabla n}{n} \frac{e}{dx} + \left(\frac{w^{2} - w^{2}}{p}\right)E = 0$$
(21)

In terms of the nondimensional variable E this becomes

$$\gamma \frac{d^2 E}{d\xi^2} - \frac{\nabla n}{n} \frac{dE}{d\xi} + \left(\frac{w^2 - w^2}{p}\right) E = 0$$
(22)

where  $\frac{\lambda_{d}^{2}}{a^{2}} = \frac{\varepsilon_{o}m}{\gamma n_{o}q^{2}} = \frac{1}{\gamma w_{o}^{2}}$  and  $w_{o}$  is the plasma frequency of the uniform plasma. Equations (21) and (22) are the perturbation equations for E. Eqs. (21) and (22) imply that the electrons move in a field of force which is due to the steady state electric field caused by the non-uniform density distribution.

#### SECTION IV

#### TRANSFORMATION OF THE PERTURBATION EQUATION

Equation (21) can be written in the form

$$\frac{d^2 E}{dx^2} - 2 f(x) \frac{dE}{dx} + g(x) E = 0$$
 (23)

where f(x) and g(x) are given by

$$f(x) = \frac{1}{2\gamma} \frac{\sqrt{n}}{\frac{n}{e}} , g(x) = \frac{w^2 - w^2}{a^2}.$$

Assume a solution to (23) of the form

$$E(x) = y(x) h(x)$$
 (24)

where h(x) is given by

$$h(x) = e^{\int_{x_0}^{x} f(x) dx}$$
(24a)

If (24) is substituted into (23), the following differential equation is obtained

$$\frac{d^2 y}{dx^2} + \Psi(x) y = 0$$
(25)

where  $\Psi(x)$  is given by

$$f(\mathbf{x}) = \frac{df}{d\mathbf{x}} + g - f^2 \quad . \tag{26}$$

When the definitions of f and g are substituted into Eq. (26),  $\Psi(x)$  becomes

$$\Psi(\mathbf{x}) = \frac{w^2 - w_p^2}{a^2} + \frac{1}{2\gamma} \frac{d^2}{dx^2} l_n \frac{w_p^2}{w_0^2} - \frac{1}{4\gamma^2} \left(\frac{w_p^2}{dx}\right)^2$$
(27)

where

$$\frac{\nabla \mathbf{u}}{\nabla \mathbf{u}} = \frac{\nabla \mathbf{w}_{s}}{\nabla \mathbf{w}_{s}} = \frac{\nabla \mathbf{w}_{s}}{\mathbf{w}_{s}} = \nabla \mathbf{v} \mathbf{u} \frac{\mathbf{w}_{s}}{\mathbf{w}_{s}}$$

Now let one nondimensionalize  $\Psi(x)$  and Eq. (25). Equation (25) becomes

$$\frac{d^2 y}{d\xi^2} + \lambda_d^2 \Psi(\mathbf{x}) \mathbf{y} = 0$$
(28)

and if one defines  $\Psi(\xi) = \lambda_d^2 \Psi(x)$ , then this becomes

$$\frac{d^2 y}{d\xi^2} + \Psi(\xi) y = 0$$
(28a)

$$\Psi(\xi) = \frac{\omega^{2} - \omega^{2}_{p}}{\gamma \omega^{2}_{o}} + \frac{1}{2\gamma} \frac{d^{2}}{d\xi^{2}} \ln \frac{\omega^{2}_{p}}{\omega^{2}_{o}} - \frac{1}{4\gamma^{2}} \left(\frac{\omega^{2}}{d\xi}\right)^{2}$$
(29)

Since  $\frac{\omega}{\omega} = e^{\theta}$ ,  $\Psi(\xi)$  can be written as  $\omega_{0}^{2}$ 

$$f(\xi) = \frac{w^2 - w^2}{\gamma w^2} + \frac{1}{2\gamma} \frac{d^2 \theta}{d \epsilon^2} - \frac{1}{4\gamma^2} \left(\frac{d\theta}{d \xi}\right)^2$$
(29a)

The integration of Eq. (17) giving  $\theta$  vs  $\xi$ , is shown in Fig. 1. The plot of  $\forall$  vs  $\xi$  for values of  $\gamma$  of  $\frac{5}{3}$  and 3 is shown in Fig. 2

If the value of f(x) is substituted into

$$\int_{e^{X}o}^{x} f(x) dx$$

$$e^{X}o = h(x),$$

it becomes

θ<u></u>

$$\int_{h(x)}^{x} f(x) dx \frac{\theta - \theta}{2\gamma} .$$
(30)

e  $2\gamma$  is just an arbitrary multiplicative constant, so that the solution for E can be given by

$$E = yh = ye^{\frac{\theta}{2\gamma}}$$
(31)

#### SECTION V

#### RESULTS

Since h(x) is a monotonically increasing function from the wall (negative  $\theta$ ) to the sheath edge ( $\theta = 0$ ), one sees that the behavior of E is determined by the behavior of y. The equation for y is given by (28a), which is recognized as a type of wave equation. If  $\Psi(\xi)$  were constant and positive, wave solutions and hence propagation of the perturbation would be possible. If  $\Psi$  were constant and negative, the solution would be a damped or growing exponential. The same type of behavior is expected,

although modified slightly, by the solutions if  $\Psi$  is a variable. Fig. 2 makes it clear that  $\Psi$  indeed does become positive for a finite range of  $\varepsilon$ .

One particularly interesting effect due to this behavior of  $\Psi$  is that the propagation is cut off near the wall where the local plasma frequency is very small. If Eq. (29a) is considered under the assumption that  $\theta$  is a large negative value, then  $\frac{w_p^2}{w_o^2} \approx 0$  and  $\frac{d^2 A}{d\xi^2} \approx 0$ , so that  $\Psi(\xi)$  becomes

$$\Psi(\xi) \approx \frac{w^2}{\gamma w^2} - \frac{1}{4\gamma^2} \left(\frac{d\theta}{d\xi}\right)^2$$
(32)

and  $\Psi(\xi)$  can then become negative for an appropriately large  $\frac{d\theta}{d\xi}$  and a cutoff can take place. Since  $\frac{d\theta}{d\xi}$  is related to the electric field, this result is not surprising in view of the fact that the electron gas is in equilibrium with the electric field  $E_0$ , and this effect is analogous to what is found in an atmosphere which is in equilibrium with a gravitational field.

In view of the complexity of Eq. (28a), this equation had to be solved numerically. In order to do this,  $\Psi(F)$  was approximated for large values of  $\xi(A\rightarrow 0)$  by a constant and the damped exponential solution for y was chosen, so that the perturbation would go to zero at infinity as it should. Then the equation was integrated backwards from this starting point at a large value of E(F=36 to be exact) to the wall. This was done for the range of frequencies of  $\frac{W}{W_0}$  from 0 to 1 and for both values of  $\gamma$ . The results for  $\gamma=3$  are shown in Fig. 3. It is seen that an oscillation does indeed occur, and also that two resonances occur.

#### SECTION VI

#### CONCLUSIONS

The results of this analysis show that the resonances in the sheath occur at frequencies below the asymptotic plasma frequency. It is of interest to observe that they occur in a relatively narrow range below  $w_0$ . Only two resonances are found, one at about  $\frac{w}{w_0} \approx 0.75$  and another at  $\frac{w}{w_0} \approx 0.95$ . In the limit of  $\frac{w}{w_0} \approx 1$ , the unperturbed plasma appears, and for larger values of  $\frac{w}{w_0}$  only a continuous spectrum of frequencies may occur. These results appear to be in agreement with the experimental behavior of the Resonance Probe.

A further examination, including the effect of collisions, is in the process of being carried out.

#### SECTION VII

#### REFERENCES

- Gabor, D., Ash. E.A., Dracott, D.: <u>Langmuirs Paradox</u>. Nature, <u>176</u>,
   p. 916, 1955.
- Von Gierke, G., Ott, W., Schwirzke, F.: <u>Untersuchung von Plasma</u>.
   <u>Grenzschichten mit eiver Electronenstrohl-Sonde</u>. Proc. Vth Int. Conf. on Ionization Phenomena in Gases, Munich, North Holland Publ. Co., Amsterdam, Vol. 2, p. 1412, 1961.
- 3. Harp, R.S. and Kino, G.S.: <u>Measurements of Fields in the Plasma Sheath by</u> <u>an Electron Beam Probing Technique</u>. Stanford University, Microwave Laboratories, Report No. 1076, September 1963.

- 4. Wimmel, H.K., <u>Theory of the Plasma Resonance Probe</u>. Z. Naturfarschg, 19a, 1964.
- 5. Pavkovich, J., Kino, G.S.: <u>Theory of the Plasma Sheath</u>, Proc. VIth Int. Conf. on Ionization Phenomena in Gases, Paris, France, Vol. 3, pp. 39-44, 1964.
- Pavkovich, J.: <u>Numerical Calculations Related to the R.F. Properties of the</u> <u>Plasma Sheath</u>, Aerospace Research Labs., Report ARL 64-17, January, 1964.
- Caruso, A. and Cavaliere, A.: <u>The Structure of the Collisionless Plasma-</u> <u>Sheath Transition.</u> IL Nuovo Cimento, <u>XXVI</u>, 6, 16 December 1962.
- Morse, R.L.: <u>Adiabatic Time Development of Plasma-Sheath</u>. Phys. Fluids,
   <u>8</u>, 2, February 1965.
- Self, S.A.: <u>Exact Solution of the Collisionless Plasma-Sheath Equation</u>. Phys. Fluids, <u>6</u>, 12, December 1963.
- Gierke, G. V., Muller, G., Peter, G., Rabben, H.H.: <u>On the Influence of</u> <u>Ion Sheaths upon the Resonance Behavior of a R.F. Plasma Probe.</u> Z. Naturforschg, 19a, 1100-1111, 1964.
- Langmuir, I. and Mott-Smith: <u>Studies of Electrical Discharges in Gases at</u>
   <u>Low Pressures.</u> General Electric Review, Part I. <u>27</u>, 1924.
- Bohm, D., Burhap, E.H.S., and Massey, H.S.W.: <u>Characteristics of</u> <u>Electrical Discharges in Magnetic Fields</u>. Chapter 2, Guthrie, A., and Wakenling, R.K. (Eds.), McGraw-Hill Book Company, New York, 1949.
- Bernstein, I. B., Robinowitz, I. N.: <u>Theory of Electrostatic Probes in a Low</u> <u>Density Plasma</u>, Phys. Fluids, <u>2</u>, 2, p. 112, 1960.
- Lam, S.H.: <u>The Langmuir Probe in a Collisionless Plasma</u>. Princeton University, Gas Dynamics Laboratory, Report 681, March 1964.

#### APPENDIX

Poisson's equation for the steady state distribution is given as

$$\frac{dE}{dx} = \frac{-q}{\epsilon_0} (n_e - n_i)$$
(A-1)

or in terms of  $\phi$  this is

$$\frac{d^2 \varphi}{dx^2} = \frac{q}{\epsilon_0} \left( n_e - n_i \right)$$
(A-2)

We know that  $n_e = n_e e_{0}^{\frac{q\varphi}{kT}}$  and we can say that  $n_i = \frac{j_i}{qu_i}$ .  $u_i$  is given by

$$u_i = \sqrt{u_i^2 - \frac{2q}{m}} v$$

where we assume that  $\varphi = 0$  at the sheath edge and  $u_i$  is the ion velocity at the sheath o edge. By the Bohm criterion for a stable sheath,  $u_i$  can be replaced by

 $u_{i_o} = \sqrt{\frac{kT_o}{m}}$ . Then Eq. (A-2) becomes

 $\frac{d^2 c}{dx^2} = \frac{q}{\epsilon_0} n_0 \left( e^{\theta} - \frac{1}{\sqrt{1-2\theta}} \right)$  (A-3)

or in terms of the nondimensional distance  $\xi$ 

$$\frac{d^2\theta}{d\xi^2} = e^{\theta} - \frac{1}{\sqrt{1-2\theta}}$$
 (A-4)

If we integrate (A - 4) and apply the boundary condition that  $\theta = 0$  at  $\frac{d\theta}{d\xi} = 0$ , then

this becomes

$$\left(\frac{\mathrm{d}\theta}{\mathrm{d}\xi}\right)^{2} = 2\mathrm{e}^{\theta} + 2\sqrt{1-2\theta} - 4 \quad . \tag{A-5}$$





FIG. 2 PROPAGATION FACTOR  $\psi$  vs DISTANCE FROM WALL



<u>Unclassified</u>				
Security Classification	A 1/4 1			
DOCUMENT CO (Security classification of tille, body of abstract and indexi	NTROL DATA - Re		the overall n	oport is classified)
1. ORIGINATING ACTIVITY (Corporate author)		24. REPOI	RT SECURIT	TY CLASSIFICATION
Polytechnic Institute of Brooklyn Dept. of Aerospace Engineering & Appli	ied Mechanics		Unclassi	
Farmingdale, New York 11735	icu neenamiee	2 b. GROUI	•	
J. REPORT TITLE	· · · · · · · · · · · · · · · · · · ·			
OSCILLATIONS IN THE PLASMA SHEATH				
4. DESCRIPTIVE HOTES (Type of report and inclusive detee) Research report	·····			
S. AUTHOR(8) (Lesi name, firei name, initial)				
Schwartz, Simon H.				
S. REPORT DATE	78. TOTAL NO. OF	PAGES	75. NO. 0	F REFS
July 1965	17			14
SA. CONTRACT OR GRANT NO.	SA. ORIGINATOR'S R	EPORT NUM	BER(S)	
Nonr 839(38)	PIBAL Rep	nort No.	880	
5. PROJECT NO.		pore ne	000	
с.	S. OTHER REPORT	NO(3) (Any	other numbe	re that may be easigned
	ente report,			
d. 10. A VA IL ABILITY/LIMITATION NOTICES				· · · · · · · · · · · · · · · · · · ·
Qualified requesters may obtain copies	s of this repo	rt from i	DDC •	
11. SUPPL EMENTARY NOTES	12. SPONSORING MILITARY ACTIVITY Office of Naval Research Electronics Branch			
	Washington	, D.C.	Page	
13. ABSTRACT				
The behavior of the ion plasma sheat sinusoidal-in-time perturbation of the p sheath is assumed, and the ion density of steady state value, with only the electric collisionless macroscopic plasma equation using an exact equation for the steady s differential equation which is linear, h obtained for the perturbation. This equ equation with variable propagation const deduced. One observes that above a cert	potential of the distribution is ron distribution ons are solved state potential but which has we uation is trans	he wall. s conside on being by a pe l distri variable sformed he analy	A cold ered to perturn rturbati bution. coeffic into the tical be	lisionless remain at the bed. The ion procedure, A second order cients is e form of a wave ehavior can be

.

.

#### Unclassified

S

ecurity	Classification	

14. KEY WORDS		LINK A		LINK B		LINKC	
		WT	ROLE	WT	ROLE	₩T	
Electron oscillations Langmuir probe Macroscopic plasma equations Plasma resonances Plasma sheath Resonance probe							
INSTRUCTIONS	•						

of the contractor, subcontractor, grantee, Department of Defense activity or other organization (corporate author) jasuing the report.

28. REPORT SECURITY CLASSIFICATION: Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulationa.

2b. GROUP: Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. REPORT TITLE: Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

DESCRIPTIVE NOTES: If appropriate, enter the type of report, e.g., interim, progress, summary, snnual, or final. Give the inclusive dates when a specific reporting period is covered.

5. AUTHOR(S): Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, abow rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. REPORT DATE: Enter the date of the report as day, month, year, or month, year. If more than one date sppears on the report, use date of publication.

7a. TOTAL NUMBER OF PAGES: The total page count ahould follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. NUMBER OF REFERENCES: Enter the total number of references cited in the report.

8a. CONTRACT OR GRANT NUMBER: If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. PROJECT NUMBER: Enter the sppropriste military department identification, such as project number, subproject number, system numbers, task number, etc.

98. ORIGINATOR'S REPORT NUMBER(S): Enter the officisl report number by which the document will be identified and controlled by the originating sctivity. This number must be unique to this report.

95. OTHER REPORT NUMBER(S): If the report has been saaigned sny other report numbers (either by the originator or by the sponsor), also enter this number(s).

10. AVAILABILITY/LIMITATION NOTICES: Enter sny limitations on further dissemination of the report, other than those

such as:

z

- (I) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not suthorIzed."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through
- (4) "U. S. military sgencies may obtain copies of this report directly from DDC. Other qualified users shall request through
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through

If the report has been furnished to the Office of Technical Services, Department of Commerce, for asle to the public, indicate this fact and enter the price, if known-

IL SUPPLEMENTARY NOTES: Use for additional explanatory notes.

12. SPONSORING MILITARY ACTIVITY: Enter the name of the departmental project office or laboratory sponsoring (paying for) the research and development. Include address

13. ABSTRACT: Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the informstion in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. KEY WORDS: Key words are technically meaningful terma or short phrases that characterize a report and may be used an index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, rales, and weights is optional.