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AD311630	
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31 Dec 1969, DoDD 5200.10; AMC D/A ltr, 4 Feb 1975	

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SCIENTIFIC AND TECHNICAL INFORMATION

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TECHNICAL INFORMATION
REPORT 6-9-6A1(3)

OFFICE, CHIEF OF ORDNANCE
DECEMBER 1957

PREPARED FOR THE U. S. ARMY
MATERIEL COMAND BY THE ARMY
MATERIEL RESEARCH STAFF,
UNIVERSITY OF PITTSBURGH,
UNDER CONTRACT DA-36-034-AMC-
3785(X)*.

DEVELOPMENT
OF

105-MM HIGH-EXPLOSIVE-PLASTIC SHELL, M327 (T81E28) (U)

Prior to our entry into World War II, it was realized that the large-caliber and more heavily armored combat vehicles being used in increasing numbers would make it necessary to furnish armor-defeating ammunition to units that heretofore had been protected by reason of their being echeloned in depth. Consequently, the development of a monobloc shot for 105-mm howitzers was approved in July 1941.

At the time authorization for this development was given, it was stated that (1) this shot should be of the same general design as that of the British 6-pounder, (2) it should not have a windshield, (3) it should have the same weight as the high-explosive (HE) shell (33 pounds), and (4) it should have a tracer. Such a projectile was designed and tested but proved to be little better than standard 75-mm armor-piercing shot against 3-inch armor plate at an obliquity of 20°. By the same authority setting up the requirement for the monobloc shot, however, work had been going on to develop a high-explosive antitank (HEAT) shell. This shell was adopted as standard in February 1942 as the M67 and retained this classification until April 1957, at which time it was replaced by the M327 (T81E28) high-explosive-plastic (HEP) shell and made limited standard.

The Ordnance Corps first became interested in HEP shell in 1947, by which time British experiments had shown the feasibility of applying this principle in their "squash-head" shell to defeat armor. Unlike kinetic-energy shot, which pierce or punch their way through armor, or HEAT shell, which use a jet of ultrahigh-velocity fragments to gain entrance, HEP shell do their damage without necessarily penetrating the plate. This is made possible by a carrier with a thin

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RELATED TIR'S

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1-57	TIR 6-9	Development of 105-mm Ammunition
4-57	TIR 6-9-5A2(4)	105-mm HEAT Shell, T131 Series
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6-56	TIR 6-9-11A3(2)	105-mm (106-mm) Canister, T310 Series

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nose that crushes upon contact, thus allowing the filler to spread over the plate just prior to detonation. The explosion of the filler sets up shock waves that spall the back of the plate, thus producing a lethal missile from the very armor that is intended for defense (see TIR E). To further investigate the possibilities of this type of ammunition, in October 1948 approval was given for a subproject under a main project, which had been authorized in October 1945, for the development of armor-defeating ammunition. This same action approved the development of the T81 shell, which was to serve as a prototype that would be capable of being fired from 105-mm howitzers and the T5E2 105-mm tank gun (the development of this gun, however, was terminated in September 1949).

In June 1949 the Office, Chief of Ordnance, furnished the contractor with a suggested design and the specifications for it. This shell, the basic T81, was to have a muzzle velocity of 1,550 feet per second when fired from a howitzer, and was to be capable of withstanding a maximum chamber pressure of 30,000 pounds per square inch. The basic components were a large-diameter base plug with a central orifice for holding a base-detonating fuze; a converted M84B1 105-mm base-ejection, smoke-shell body (with the walls thinned down and the nose cut off); and a thin, drawn, ogival steel nose. The plug was screwed into the base of the projectile, into whose opposite end the nose was butt-welded. This design was agreed upon in July 1949 by representatives of Picatinny Arsenal and the contractor, who also concurred in the opinion that the investigation of the T81 should concern itself with the shape, length, material, thickness, and the hardness of the nose; also, to be considered were methods of affixing the nose and body together and the means of attaching a windshield if such an item were to be used (see accompanying chart). In addition, it was deemed advisable to make each modification differ from the basic design with respect to only one of the features listed above and to make the modifications in pairs with each of the modified designs varying from the T81 in different respects, such as a longer or a shorter nose and a longer or a shorter body.

In April 1950, the first firing tests of shell with the basic design were held at Aberdeen Proving Ground. The results showed that this two-piece shell could defeat 6 inches of brittle armor with a Charpy value of 8 foot-pounds, but not an equivalent thickness of rough armor having a Charpy value of 52 foot-pounds. It was felt, however, that the test was a success because it demonstrated the desirability of devoting additional work to the development of HEP shell. All of the subsequent shell of the T81 series, up to and including the T81E17, were two-piece models made from converted smoke shell and were designed to meet the same requirements that were established for the T81.

During the course of development, it was found that, because of its ductility, the nose of the two-piece shell tended to bulge in the region adjacent to the brazed joint when fired at muzzle velocities of about 1,300 feet per second. In an effort to find a solution to this problem, a meeting was held in October 1950, at which time it was decided that one-piece shell should be fabricated. The most practicable method of manufacture, it was agreed, was to draw the shell body and nose in one piece to form the internal contour, and then spin the nose shut to form the ogive. In order to conserve time and

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materials, the initial work was done on the T165E11 75-mm shell, which was being developed for light tank guns. Tools and processes that proved successful were to be modified, as needed, for use in the production of 105-mm models.

After some initial one-piece shell had been fired, it was decided that such a design would be advantageous for all calibers, not only because its terminal-ballistic characteristics appeared to be at least as good as those of the best two-piece shell and its muzzle velocity higher, but also because it would cost less to produce the one-piece item; it could be fabricated with a greater degree of uniformity; the equipment on hand could be used to better advantage; and there was a larger supply of material available for its manufacture than was on hand for two-piece shell.

By November 1951 efforts were being made, wherever possible, to replace two-piece shell with the one-piece kind. The required performance characteristics for the T81E18 were as follows:

1. Chamber pressure - 30,000 pounds per square inch
2. Design pressure - 33,600 pounds per square inch
(112 per cent of operating pressure)
3. Muzzle velocity - 2,000 feet per second
4. Terminal effect - the ability to defeat a minimum of 5 inches of armor plate (35 to 40 Charpy) at obliquities of 0° through 60°
5. Accuracy - comparable to that of the HE shell up to a range of 2,000 yards

The T81E18, having the same general shape as the T81E17, had an ogival nose and a square base, but, in place of a conventional, recessed base plug, it had a flanged plug with a flat copper gasket. This type of base closure had been tried on the 75-mm shell and proved to be as effective as well as an economical seal. Ballistically, this shell was to match the T131 105-mm HEAT shell, which was designed to have a muzzle velocity of 2,000 feet per second. Tests of the T81E18 indicated that it could spall 5-inch plate at 60° at striking velocities varying from 1,725 to 2,025 feet per second; no spalling occurred at striking velocities below 1,650 feet per second. Its probable error was 0.19 mil vertically and 0.50 mil horizontally at 1,000 yards when fired at a muzzle velocity of 2,050 feet per second as compared to 0.32 mil and 0.62 mil, respectively, for the M1 105-mm HE shell with a muzzle velocity of 1,550 feet per second.

When it was found that the T131 HEAT shell could not withstand a muzzle velocity of 2,000 feet per second, the requirement was lowered

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DESIGN SUMMARY

DESIGN	DESCRIPTION	REASON FOR CONSIDERATION										REMARKS (Numbers below correspond to those in 140)
		1	2	3	4	5	6	7	8	9	10	
		ROCK MATERIAL	ROCK SHAPE	ROCK LENGTH	ROCK THICKNESS	ROCK THICKNESS	CONCENTRATION OF FALLOUT	SHELL LENGTH	EXPANSION OF CORE	LENGTH OF NOSE ANNEAL	OTHER USE REMARKS	
T01	An M401 105-mm smoke shell body fitted with an annealed, 1010 steel, 12 (length), 15.35 inches; as-fired weight, 20.3 pounds; nose length, 5.2 inches.	X	X	X								1. Tests against armor plate indicated that the shell was better than the later models had steel noses. 2. Tests indicated that the ogival nose was better than the hemispherical or convex nose. 3. A true comparison could not be made because of distribution firing conditions.
T01E1	Like the basic T01 but with a 0.113-inch thick, annealed copper, ogival shaped nose.	X										1. Tests against armor plate indicated that the shell was better than the later models had steel noses. 2. Tests indicated that a hemispherical nose was inferior to an ogival nose. 3. There was a slight indication that, in general, the copper nose was superior to the armor.
T01E2	Like the basic T01 except for a hemispherical nose - over-all length, 23 inches; as-fired weight, 23 pounds.		X	X								1. Tests indicated that a hemispherical nose was inferior to an ogival nose. 2. There was a slight indication that, in general, the copper nose was superior to the armor.
T01E3	Like the T01 except for a convex nose - over-all length, 15.35 inches; as-fired weight, 23.5 pounds.		X									1. Tests indicated that a convex nose was inferior to an ogival nose.
T01E4	Like the basic T01 except for a shorter nose (4.0 inches) and a longer body to give same over-all length of 15.35 inches - as-fired weight, 23.5 pounds.			X								1. There was no apparent difference in the results. This shell was not consistent in spalling 3, 4, 5, or 6-inch armor.
T01E5	Like the basic T01 except the nose length is 6 inches - over-all length, 15.35 inches; as-fired weight, 21.3 pounds.			X	X							1. There was no apparent difference in the results. This shell was not consistent in spalling 3, 4, 5, or 6-inch armor. 2. There was apparently no large difference in contact area at 60° obliquity. 3. Apparently, the T01E5 was inferior to others. It was concluded that a shell with a nose length of 6 inches was better than a shell with a nose length of 4 inches. It was also concluded that the more distorted the shell, the more distorted the results of the ogive to withstand filler pressure.
T01E6	Like the basic T01 except the nose length is 8 inches - over-all length, 15.35 inches; as-fired weight, 22.5 pounds.			X								1. There was no apparent difference in the results. This shell was not consistent in spalling 3, 4, 5, or 6-inch armor.
T01E7	Like the T01E2 except it has a ductile steel windshield based on the nose - over-all length, 22.30 inches.		X									2. No tests were conducted inasmuch as impact tests showed that the ductile steel windshield was not likely to pull up between the filler and the armor.

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DESIGN	DESCRIPTION	REASON FOR CONSIDERATION								REMARKS (Numbers below correspond to those in left)		
		1	2	3	4	5	6	7	8		9	10
		MORE MATERIAL	MORE SHAPE	MORE LENGTH	MORE WIDENESS	MORE THICKNESS	COMPARISON OF FILLERS	SHELL LENGTH	COMPARISON OF CORE		CARET OF MISE AREA	OTHER USE (SEE REMARKS)
T01E1	Like the T01E1 except that the ogive is "sand-drums" over-all length, 15.55 inches; as-fired weight, 31.5 pounds.				X						4. There was apparently no large difference in contact areas at GP alignment.	
T01E2	Like the T01E1 except that the ogive is cone-hardened to a depth of 0.003 to all lengths, 15.55 inches; as-fired weight, 31.5 pounds.				X						4. There was apparently no large difference in contact areas at GP alignment.	
T01E3	Like the T01E1 except it has a 7-degree as-fired weight, 31.5 pounds.					X					5. It was apparent that the T01E3 was superior to the T01E2 but inferior to the T01E1. However, ballistic tests indicated that the T01E3 was superior to a T01E2 made up in prior to a thin nose.	
T01E4	Like the T01E1 except it has a 15-degree ogive - over-all length, 15.55 inches; as-fired weight, 31.5 pounds.					X					5. It was apparent that the T01E4 was superior to both the T01E2 and T01E3. However, ballistic tests indicated that the T01E4 was superior to a T01E2 made up in prior to a thin nose. Increased the ability of the ogive to withstand filler pressure.	
T01E5	Like the T01E1 except it has a cast on the nose - over-all length, 21.79 inches; as-fired weight, 31.5 pounds.	X									5. No firing tests were conducted because as tests with shell having conical noses and no stabilizing proved that the T01E5 was superior to a T01E2 made up in prior to a thin nose.	
T01E6	Like the T01E1 but with internal grooves cut in the body, and a wedge ring between the body and the ogive to facilitate peeling action.									X	5b. This relatively different design was used to determine if the T01E6 was superior to the T01E5. The results of the firing tests were negative. The firing tests were made because of the difficulty of making the groove, and sheets of filler pressure were observed. The filler was too widely dispersed.	
T01E7	Like the T01E1 except the hemispherical ogive - over-all length, 15.55 inches.			X							5. None of these shells were fired.	
T01E8	Like the T01E1 except the nose is 6-inches long - over-all length, 17.55 inches; as-fired weight, 31.5 pounds.	X									2. This shell seemed to be the best one of the series. It was felt, however, that it was not ideal. Therefore, it was decided to make a series of shells designed to be better than that of the T01E1.	
T01E9	Like the T01E1 except the nose is 6-inches long - over-all length, 17.55 inches; as-fired weight, 31.5 pounds.	X									3. There was a slight indication that the longer nose was generally superior to the shorter nose. However, it was subject to subsequent testing.	

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DESIGN SUMMARY

DESIGN	DESCRIPTION	REASON FOR CONSIDERATION										REMARKS (Numbers below correspond to those to left)
		1	2	3	4	5	6	7	8	9	10	
		NOSE MATERIAL	NOSE SHAPE	NOSE LENGTH	NOSE THICKNESS	THICKNESS	COMPARISON OF PALLETS	SHELL LENGTH	IMPEDANCE CONE	LENGTH OF NOSE JACKET	OTHER (SEE REMARKS)	
T01E17	Similar to the T01E16 except the nose is shorter, 12.5 inches; 44-fired weight, 22.4 pounds.		X				X					2. This shell proved to be the equivalent of the T01E16 and, with the exception of the slightly different aspect to terminal effect. It was shown that the T01E17 was superior, from the standpoint of exterior ballistics.
T01E18	Same general shape as the T01E17 but with a flanged-type base plug and a flat copper jacket - over-all length 16.3 inches; 44-fired weight, approximately 21.9 pounds.										X	3. This shell was designed to be a ballistic match with the T01E16 and at a muzzle velocity of 2,000 f.p.s. It was shown, however, after the AT competition, that the T01E18 was superior in spalling effect.
T01E19	Like the T01E18 except the ogive is smaller - over-all length, 16.3 inches; 44-fired weight, 21.5 pounds.										X	4. This shell was designed to be effective against the T01E16. Some spalled 5-inch plate at striking velocity of 1,745 to 1,770 f.p.s. The T01E19 was successful in meeting the required terminal effect when tested with Competition 11.
T01E20	Like the T01E17 but with a thicker (6-gage) nose - over-all length, 17.56 inches.										X	5. This projectile was designed to investigate the possibilities of having a two-piece shell that would not deform upon impact. It showed evidence of deformation upon firing and could not be relied upon to split 5-inch plate.
T01E21	Like the T01E19 except for a longer (12-inch) ogive and shorter body to achieve the same over-all length of 16.3 inches; 44-fired weight, 24.3 pounds.										X	6. This shell was designed as a precaution in the event that the one-piece T01E19 proved unsatisfactory. Some deformation was observed in the one-piece shell was developed.
T01E22	Similar to the T01E20 except the base-cup section of the converted nose is shorter - over-all length, 15.52 inches; 44-fired weight, 24.3 pounds; ogive length, 10.3 inches.										X	7. This shell was designed to increase the strength and stability of the T01E20. It was considered, however, because of the T01E19 heavy-weighted shell.
T01E23	Design calls for cutting the T01E19 shell in half, machining a top shoulder on the front of the nose half, and the rear of the shell half, and then joining the two halves together at a spot just forward of the rear band - over-all length, 16.04 inches.										X	8. This shell was designed as the result of the poor showing of the T01E19. It was shown that the T01E23 would be satisfactory.

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DESIGN SUMMARY

DESIGN	DESCRIPTION	REASON FOR CONSIDERATION										REMARKS (Numbers below correspond to those to left)
		1	2	3	4	5	6	7	8	9	10	
		ROSE MATERIAL	ROSE SHAPE	ROSE LENGTH	ROSE WIDTH	ROSE THICKNESS	COMPARISON OF FILLERS	SHELL LENGTH	COMPRESSION OF CORE	LENGTH OF ROSE ANNEAL	OTHER (SEE REMARKS)	
TWE21	Like the TWEL18 except for a dispersion cone in the nose.								X			9. No design study was made.
TWE22	Like the TWEL20 except for a dispersion cone in the nose.								X			9. No design study was made.
TWE23	Like the TWEL18 except that it is 1-1/2 inches longer - over-all length, 11.13 inches.							X				7. No test results are available.
TWE24	Like the TWEL18 except it is annealed back to 5 inches from the tip of the nose.									X		9. These Composition C4 filled shells were effective against 5-inch armor plate at a striking velocity of 1,475 to 1,525 fps. It appeared to be the equivalent of the TWEL25.
TWE25	Like the TWEL18 except it is annealed back to 7 inches from the tip of the nose.									X		9. This shell, when filled with Composition C4, could defeat 5-inch armor plate at a striking velocity of 1,475 to 1,525 fps. When filled with Composition A1, it could defeat the same armor at velocities of 1,375 to 1,525 fps.
TWE26	Like the TWEL18 except it is one inch longer and annealed for 5 inches back from the tip of the nose.									X		9. This shell was as effective against 5-inch armor as the TWEL25 and TWEL26. When Composition A1 loaded the shell could defeat armor plate at a striking velocity of 1,375 fps.
TWE27	Like the TWEL18 except it is 1-inch longer.							X				7. This shell, when loaded with Composition C4, was effective at striking velocities of from 1,175 to 1,475 fps. (C4 was used in the test.) At the same striking velocities when loaded with Composition A1, it was effective against armor plate as the TWEL25, TWEL26, or the TWEL28.
TWE28 (B377)	The TWEL18A with a narrower slit copper gasket set in a recess in the base of the shell - loaded with Composition A1.										X	20. This shell was formerly the TWEL18A, which was developed to be effective against armor plate. When filled with Composition A1, it was found to be effective against armor plate at velocities between 1,375 and 1,475 fps. There is no longer a need for this shell. The way it was made standardized at a muzzle velocity of 1,000 fps.

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to 1,800 feet per second. This meant that, if the two shell were to match ballistically, the velocity of the T81E18 would also have to be lowered. This was not feasible, however, since it would result in too small a spread between the muzzle velocity of 1,800 feet per second and the minimum effective striking velocity of 1,650 feet per second; this spread, in fact, was so small that it limited the effective range to about 200 yards. In an attempt to meet the requirement, the design was changed to include an annealed nose. This shell - designated the T81E18A - failed to spall 5-inch armor at 0° and 60° obliquity when striking at velocities of from 1,275 to 1,775 feet per second. As a result, consideration was given to the T81E19 and T81E20 two-piece shell and modifications of the latter (designs A, B, and C). Only the T81E19 was manufactured, and it proved to be unable to spall 5-inch armor at 55°.

To determine whether a dispersion cone might aid in increasing terminal effect, two designs - the T81E21 and the T81E22 - were considered. The T81E21 was to be like the T81E18 and the T81E22 was to be similar to the T81E20, but the two new designs called for a dispersion cone in the nose of each. No design study was made, however, so that no tests were run.

The additional models that followed the T81E22, up to and including the T81E28, were for the purpose of testing the effect of body length and nose length on terminal-ballistic performance. In addition, the noses of some of these models were to be annealed to various distances from their tips to see if terminal effectiveness was changed in any way thereby. The designation of the T81E18A was changed to T81E28 because it was felt advisable to assign new designations when the heat treatment used in the fabrication of any projectile was varied. A slight additional change was the employment of a narrower flat copper gasket, which was set in a recess in the shell base. It was found that, when loaded with Composition A3 in place of Composition C4, this projectile demonstrated a satisfactory terminal effectiveness at a striking velocity as low as 1,375 feet per second. Following user tests during the latter half of 1955, Continental Army Command (CONARC) recommended that the T81E28, having a muzzle velocity of 1,900 feet per second, be adopted as standard. This was approved by the Ordnance Technical Committee in April 1957, at which time the T81E28 was designated the M327. A muzzle velocity of 1,900 feet per second was chosen for the T81E28, because velocities higher than that affect stability adversely. It is believed that by redesigning the rotating band and by using inert nose pads developed in 1957 to prevent shock prior to the functioning of the fuze; it will be possible to fire the shell at higher velocities and improve the probability of a first-round hit. However, nothing has been done along these lines since there is now no requirement for the continued development of this shell.

Like the majority of HEP shell, the M327 differs structurally from conventional artillery shell by having a thin forward wall and ogive and by being loaded from the base end. A threaded steel base plug with a centrally threaded orifice for holding an M91A1 base-detonating fuze closes off the rear of the projectile. The filler, 7.6 pounds of Composition A3, is further protected from the propelling gases by a flat copper gasket that seals the jointure between the

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CROSS SECTION OF 105-MM HEP SHELL, M327 (T81E28)

- | | |
|--------------------------|------------------|
| 1. Cartridge case | 6. Copper gasket |
| 2. Propelling-charge bag | 7. Felt washer |
| 3. Propellant | 8. Rotating band |
| 4. Primer | 9. Fuze |
| 5. Tracer | 10. Felt disk |
| 11. Explosive charge | |

shoulder of the base plug and the rim of the body. The fuze is sealed by a concentric, copper-backed, lead caulking ring. A pressed felt washer, between the filler and the face of the base plug, and a pressed felt disk, between the forward face of the fuze and the filler, reduce the shock imparted to the filler by setback. A single gilding-metal rotating band is pressed into a groove about the base of the projectile to complete the assembly.

A complete M327 round is assembled as a semifixed round consisting of the projectile, an M14B1 steel cartridge case, a single bagged propelling charge containing 58.15 ounces of M6 MP propellant, and an M28B2 percussion primer.

Terminal effectiveness tests have shown that 80 per cent of the hits will defeat 5-inch rolled homogeneous plate (of 35 to 50 foot-pound Charpy value at -40° F and a Brinell hardness number varying from about 225 to 262), when striking at obliquities ranging from 0° to 60°. The remaining 20 per cent of the hits will produce hinged spalls or bulges on the rear face of the plate.

When the M327 was adopted as standard, it was stated that user tests of the T131E31 105-mm HEAT shell indicated that it may be more effective than the HEP shell for the defeat of armor but that its present degree of accuracy is unacceptable. In keeping with CONARC's

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findings and proposals, quantity procurement of the M327 HEP shell is being held in abeyance pending the outcome of the development of the T131 series.

The following characteristics are for the M327 (T81E28) round only.

PRINCIPAL CHARACTERISTICS

Caliber	105 mm
Models of weapon in which used	
Cannon for SP howitzers	M2A2, M4, M4A1, M49, T252
Field cannon	M2A1, M2A2
Projectile	
Weight, as fired	23.38 lb
Length with fuze	17.06 in
Charge	Comp A3
Weight	7.6 lb
Stabilization	spin
Fuze	M91A1 BD
Cartridge case	M14B1
Length	14.64 in
Weight	5.9 lb
Propellant	M6
Weight	3.64 lb
Primer	M28B2
Length of complete round	29.08 in
Weight of complete round	33.45 lb
Performance	
Spalling of homogeneous armor	
1,000 yd	5 in
2,000 yd	5 in
Probability of hit	
1,000 yd	
With range finder	0.98
Without range finder	0.31
2,000 yd	
With range finder	0.17
Without range finder	0.06
Probable error (H and V)	0.2 mil
Maximum tactical range (against armor)	2,000 yd
Muzzle velocity	1,900 fps

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