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Attorney Docket No. 84493
Date: 4 January 2005

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Serial Number: 11/015,805

Filing Date: 20 December 2005

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DISTRIBUTION STATEMENT A
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20060111 052

1 Attorney Docket No. 84493

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METHOD FOR DETERMINING LOCAL INNER AND OUTER

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BOUNDARY LAYER LENGTH SCALES FROM DRAG MEASUREMENTS

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IN HIGH REYNOLDS NUMBER TURBULENT FLOWS

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STATEMENT OF GOVERNMENT INTEREST

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The invention described herein may be manufactured and used

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by or for the Government of the United States of America for

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governmental purposes without the payment of any royalties

11

thereon or therefore.

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13

BACKGROUND OF THE INVENTION

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1. Field of the Invention

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The present invention generally relates to the estimation

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of the flow noise on a cylindrical body in a turbulent flow and,

17

more particularly, to a method for determining inner and outer

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boundary layer length scales from a succession of drag

19

measurements of a long thin cylindrical body in any fluid as a

20

precursor to estimating flow noise.

21

2. Description of the Prior Art

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There is a significant practical need to know the drag and

23

flow noise of towed long thin cylindrical bodies. The need

1 arises in a variety of contexts including torpedoes and towed
2 sonar arrays.

3 Towed sonar arrays are sonar systems that are designed to
4 be towed by a submarine or a surface vessel in order to detect
5 other submarines. The arrays are typically long, hose-like
6 structures measuring up to a thousand feet or longer that
7 contain specially designed acoustic sensors, called hydrophones,
8 which receive acoustic waves. The arrays include electronics
9 that convert the acoustical waves from analog to digital form
10 and transmit that data to electronic processors on board the
11 towing vessel.

12 The processor must distinguish radiated sound from other
13 submarines from ambient and self noise, which includes the flow
14 noise of the towed array. Thus, it is important to accurately
15 estimate flow noise in advance, for design purposes. Moreover,
16 towed arrays must be designed to withstand the extreme
17 environmental stresses of operation in the ocean depths, and so
18 it is necessary to accurately estimate drag, and estimate the
19 local wall shear stress as well. Accomplishing this requires an
20 understanding of the turbulent boundary layers which exist on
21 the arrays.

22 The inner region of the boundary layer is dominated by
23 viscous effects, and the outer region is dominated by inertial
24 effects. Two dimensional flat plate turbulent boundary layers

1 have been explored thoroughly for several decades, and it is
2 generally accepted that the (inner) viscous length scale and the
3 (outer) boundary layer momentum thickness scale adequately
4 characterize the flow.

5 Most practical engineering flows, however, are
6 characterized as high-Reynolds number flows. Since the viscous
7 length scale decreases rapidly with increasing Reynolds number,
8 and the outer length scales are only a weak function of Reynolds
9 number, the inner and outer scales become increasingly disparate
10 with increasing Reynolds number. Thus, more complex turbulent
11 flows are often not well described by the Reynolds number alone,
12 and must be described using inner and outer boundary layer
13 length scales.

14 In the context of a towed array, the hydrodynamic flow is
15 a high Reynolds number turbulent boundary layer, which may be
16 equilibrium or nonequilibrium depending on the ship motion.
17 Consequently, it is necessary to know the inner and outer
18 boundary layer length scales, which characterize the flow field,
19 for estimation of flow noise on long thin cylinders, and in
20 particular, current and next generation towed sonar arrays.

21 Currently there are no viable approaches for determining
22 the inner and outer boundary layer length scales in tow tank
23 testing or full scale sea trials. Laser Doppler Velocimetry
24 (LDV) and Particle Image Velocimetry (PIV) have been used

1 extensively for measurements of turbulence in laboratories.
2 However, oceanic field applications are impractical. It would
3 be greatly advantageous to provide a method for determining
4 inner and outer boundary layer length scales and, more
5 particularly, from a succession of drag measurements of a long
6 thin cylindrical body, in order to estimate flow noise and for
7 improved computational modeling of the dynamics of towed arrays
8 in water or other towed bodies in air.

9

10

SUMMARY OF THE INVENTION

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Accordingly, it is an object of the present invention to
12 provide a method for determining inner and outer boundary layer
13 length scales.

14

It is another object of the present invention to provide a
15 method for determining inner and outer boundary layer length
16 scales from a succession of drag measurements of a long thin
17 cylindrical body.

18

It is still another object of the present invention to
19 provide a method for determining inner and outer boundary layer
20 length scales of a long thin cylindrical body in order to
21 estimate flow noise and for improved computational modeling of
22 the dynamics of towed arrays in water.

23

In accordance with the stated objects, a method is provided
24 for determining the local inner and outer turbulent boundary

1 layer length scales from experimental measurements of the drag
2 on a long thin cylindrical body at low or high momentum
3 thickness Reynolds numbers. A succession of measurements of the
4 total drag on a cylinder under tow is taken for particular
5 conditions (flow speed, fluid density, fluid viscosity,
6 cylindrical body geometry). After each measurement the cylinder
7 is truncated by a fixed amount, and the process is repeated for
8 the entire length of the cylinder. The collective measurements
9 provide a spatially and temporally averaged measure of the mean
10 wall shear stress and momentum thickness, from which the inner
11 and outer length scales can be determined directly, for each
12 separate segment of the cylinder. The inner and outer boundary
13 layer length scales may then be used for estimation of flow
14 noise on long thin cylinders, and in particular, current and
15 next generation towed sonar arrays. In particular, this method
16 also allows the spatial variation of the length scales down the
17 length of the cylinder to be determined directly.

18 The present invention reduces the time and overhead
19 required to produce the accurate flow data needed for proper
20 engineering of towed sonar arrays.

1
2 **BRIEF DESCRIPTION OF THE DRAWINGS**

3 FIG. 1 is a schematic drawing illustrating the towing
4 configuration and load cell used in accordance with the present
5 method;

6 FIG. 2 is a diagram of the control volume for cylindrical
7 coordinates based on a side view of a tested cylinder; and

8 FIG. 3 is an end view of the control volume for cylindrical
9 coordinates from the view of reference line 3-3 of FIG.2.

10 **DETAILED DESCRIPTION OF THE PREFERRED EMBODIMENTS**

11 The present invention is a method for determining inner and
12 outer boundary layer length scales from a succession of drag
13 measurements of a long thin cylindrical body in any fluid as a
14 precursor to estimating drag and flow noise.

15 The methodology begins by towing a unit under test (UUT),
16 preferably a long thin neutrally buoyant cylinder 10, in a
17 controlled environment such as a towing tank, or from a surface
18 platform under conditions for which the ambient flow field is
19 known.

20 FIG. 1 is a schematic drawing illustrating the requisite
21 towing configuration, which includes a movable vehicle capable
22 of towing the UUT 10. The vehicle may be any air or sea vessel
23 or, as illustrated in FIG. 1, a movable tow platform 12 capable
24 of towing the UUT 10 in the illustrated tow direction "A"

1 through a fluid medium 40 (here illustrated as water). In the
2 illustrated embodiment the tow carriage 12 is mounted over a tow
3 tank. A processor 14 such as a conventional laptop computer is
4 supplied, here on the tow carriage 12, and is coupled for data
5 transmission (by RS-232, USB port or otherwise) to a load cell
6 computer interface 16. One skilled in the art will recognize
7 that the processor 14 may be any suitable computer located on-
8 site or in remote communication with the load cell computer
9 interface 16.

10 The load cell computer interface 16 may be a conventional
11 multi-meter as will be described or any other interface capable
12 of digitizing the analog voltage signal produced by a load cell
13 22. A conventional fixed tow strut 18 extends beneath the tow
14 carriage 12 into the fluid medium 40. The load cell 22 is
15 mounted distally on the tow strut 18 behind a common fairing 24
16 which minimizes the generation of turbulence. The load cell 22
17 may be any common type of tensile load measuring device, such as
18 a strain gage load cell. An axial type load cell usually
19 consists of a hollow or solid cylindrical shaft and four strain
20 gages mounted around the circumference.

21 The strain gages are mounted and connected to form a
22 Wheatstone bridge circuit. The load cell 22 is tethered by a
23 leader line 26 to the UUT 10, which is depicted as a small

1 diameter cylinder. The leader line 26 separates the UUT 10 from
2 any turbulence generated by the tow strut 18.

3 In practicing the method of the invention, the UUT 10 is
4 towed and the total drag on the towed cylinder is measured
5 directly by the load cell 22, which outputs an analog signal
6 that is digitized by the load cell computer interface 16. The
7 digitized load is processed using a control volume analysis
8 extended to the case of axisymmetric flows to exactly calculate
9 the momentum thickness (which is the outer length scale) of the
10 turbulent boundary layer at the end of the cylinder UUT 10.

11 A suitable control volume analysis is detailed below in
12 reference to FIGS. 2 and 3. This calculation requires the angle
13 of tow of the UUT 10 to be within one degree, and the tow speed
14 U_0 to be steady temporally.

15 Next, a fixed-length segment of the UUT 10 is removed from
16 its trailing end, and the total drag on the towed cylinder is
17 measured directly as described above by the load cell 22, and
18 the control volume analysis of axisymmetric flow is repeated to
19 calculate the momentum thickness of the truncated cylinder UUT.

20 The foregoing procedure is repeated successively, with a
21 fixed segment of the cylinder UUT 10 being removed for each drag
22 measurement. A typical UUT segment length to remove is
23 approximately 1 m, but could be larger or smaller, depending on

1 the desired spatial resolution. The foregoing procedure is
2 repeated for the entire length of the UUT 10.

3 It can be seen that the difference in drag between
4 consecutive measurements yields the spatially and temporally
5 averaged mean wall shear that exists on each particular segment.
6 By repeating this procedure over the entire length of the UUT
7 10, the spatial dependence of the mean wall shear stress is
8 determined, as well as the spatial dependence of the momentum
9 thickness.

10 FIG. 2 is a diagram of the control volume for cylindrical
11 coordinates. A standard control volume analysis is applied in
12 which the following notations are used.

13 a cylinder radius (ft)

14 $u(r)$ temporal mean streamwise velocity at radial
15 location r (ft/sec)

16 r radial distance from the center of the cylinder
17 (ft)

18 x streamwise distance from the leading edge of the
19 cylinder (ft)

20 U_0 tow speed of the cylinder (ft/sec)

21 CS control surface of the control volume (ft²)

22 dA incremental annular surface area at the end of
23 the control volume (ft²)

1 \bar{F} vector force applied to the surface of the
2 control volume (lbf)
3 F streamwise scalar force applied to the surface of
4 the control volume (lbf)
5 A_s total surface area of the cylinder (ft²)
6 A_2 annular surface area at the end of the control
7 volume (ft²)
8 L length of the cylinder (ft)
9 C_d tangential drag coefficient (nondimensional)
10 s boundary layer inner length scale (ft)
11 ν kinematic viscosity of the fluid (ft²/sec)
12 μ_t friction velocity (ft/sec)
13 τ_w temporally averaged mean wall shear stress (psf)
14 τ_{ave} spatially and temporally averaged mean wall shear
15 stress (psf)
16 ρ fluid density (slugs/ft³)
17 δ boundary layer thickness at the end of the
18 cylinder (ft)
19 boundary layer momentum thickness at the end of the
20 cylinder (ft)
21 \bar{V} temporal mean velocity vector (ft/sec)
22 d connotes the derivative of the associated term

1 The outer boundary layer length scale is the momentum
2 thickness δ itself, and the inner boundary layer length scale
3 is given by ν/μ_τ , where $\mu_\tau = (\tau_w/\rho)^{1/2}$.

4 Using cylindrical coordinates, as shown in FIGS. 2 and 3,
5 an expression for the momentum thickness δ is derived. For
6 convenience, the radius of the cylindrical control volume is
7 chosen to be equal to the value of the boundary layer thickness δ
8 at the end of the cylinder, and the length is the total length
9 of the cylinder or UUT 10. The origin is the centerline 120 of
10 the cylinder, such that the surface of the cylinder is at $r = a$.
11 In the following analysis, all quantities represent the temporal
12 mean values. We first present an expression defining the
13 momentum thickness δ for this case. Equating the momentum flux
14 through an annulus in the free stream, to the momentum flux
15 defect in the boundary layer, leads to

16
$$\delta^2 + 2a\delta = 2 \int_a^{\delta+a} \frac{u(r)}{U_o} \left(1 - \frac{u(r)}{U_o} \right) r dr \quad (1)$$

17 which again can only be evaluated for δ , if $u(r)$ the mean
18 streamwise velocity in the boundary layer is known. However,
19 the momentum thickness may also be derived using a control
20 volume analysis. A cylindrical control volume is used, as shown
21 in FIG. 2, and steady-state conditions are imposed.
22 Conservation of mass for the control volume yields

1
$$\int_{CS} \rho \bar{V}(r, \theta, x) \cdot d\bar{A} = 0 \quad (2)$$

2 where vector \bar{V} is the temporal mean velocity at the location of
3 the control volume surfaces. Conservation of momentum applied to
4 the control volume can be written as

5
$$\int_{CS} \bar{V} \rho \bar{V} \cdot d\bar{A} = \sum \bar{F} \quad (3)$$

6 Note that the only applied force \bar{F} on the cylindrical control
7 volume of fluid is the shear force at the wall of the cylinder.
8 This force is equal to the streamwise component of the mean wall
9 shear stress averaged over the surface area of the entire
10 cylinder multiplied by the total surface area $A_s = 2\pi aL$.
11 Evaluating the integral at each control surface, and making use
12 of equation (2), yields

13
$$\frac{\tau_{ave} A_s}{\rho U_o^2} = \int_{A_2} \frac{u(r)}{U_o} \left(1 - \frac{u(r)}{U_o} \right) dA \quad (4)$$

14 where $dA = r dr d\theta$. The quantity τ_{ave} which is inferred from the
15 drag measurements, is related to the spatially varying wall
16 shear stress through the relation

17
$$\tau_{ave} = \frac{1}{L} \int_0^L \tau_w(x) dx \quad (5)$$

1 Equation (4) can be simplified to

$$2 \quad \frac{\tau_{ave}}{\rho U_o^2} = \frac{1}{L} \int_a^{a+\delta} \frac{u(r)}{U_o} \left(1 - \frac{u(r)}{U_o}\right) \frac{r}{a} dr = \frac{1}{2} C_d \quad (6)$$

3 where C_d is the total tangential drag coefficient over the
4 cylinder length L .

5 Thus, from the measured quantity C_d , equation (6) can be solved
6 for the temporally and spatially averaged mean wall shear stress
7 τ_{ave} existing on each segment of the cylindrical body. From τ_{ave} ,
8 the inner boundary layer length scale s can be directly
9 determined.

10 Using equation (1) for the definition of momentum thickness
11 in conjunction with the control volume analysis, the following
12 relationship is obtained between θ evaluated at $x = L$ and C_d for
13 the case of a cylinder in a steady, uniform flow:

$$14 \quad \theta^2 + 2a\theta - aLC_d = 0 \quad (7)$$

15 The outer boundary layer length scale is the momentum
16 thickness θ itself, which is determined directly from equation
17 (7), with the measured value of C_d known.

18 Thus, we now have an accurate determination of the inner
19 and outer boundary layer length scales s and θ (the inner viscous
20 length scale and the outer boundary layer momentum thickness
21 scale), which are generally recognized as adequately

1 characterizing the flow. The calculations are derived very
2 simply from a succession of drag measurements of a long thin
3 cylindrical body. The calculations may then be used for the
4 estimation of flow noise and improved computational modeling of
5 the dynamics of towed bodies in fluids such as air or water.
6 This greatly reduces the time and overhead required to produce
7 the accurate flow data needed for proper engineering of towed
8 sonar arrays.

9 Having now fully set forth the preferred embodiments and
10 certain modifications of the concept underlying the present
11 invention, various other embodiments as well as certain
12 variations and modifications of the embodiments herein shown and
13 described will obviously occur to those skilled in the art upon
14 becoming familiar with said underlying concept. It is to be
15 understood, therefore, that the invention may be practiced
16 otherwise than as specifically set forth in the following
17 claims.

1 Attorney Docket No. 84493

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METHOD FOR DETERMINING LOCAL INNER AND OUTER

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BOUNDARY LAYER LENGTH SCALES FROM DRAG MEASUREMENTS

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IN HIGH REYNOLDS NUMBER TURBULENT FLOWS

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ABSTRACT OF THE DISCLOSURE

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A method is presented for determining inner and outer

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boundary layer length scales from a succession of drag

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measurements of a cylindrical body in order to estimate flow

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noise and for computational modeling of the dynamics of towed

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arrays in a fluid medium. A succession of measurements of the

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total drag on a cylinder under tow at uniform known conditions

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(flow speed, fluid density, fluid viscosity, cylindrical body

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geometry) is taken. After each measurement, the cylinder is

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truncated by a fixed amount, and the process is repeated for the

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length of the cylinder. The measurements provide a spatially

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and temporally averaged measure of the mean wall shear stress

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and momentum thickness, from which the inner and outer length

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scales can be determined. The inner and outer boundary layer

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length scales may then be used for estimation of flow noise on

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towed cylindrical bodies and arrays.

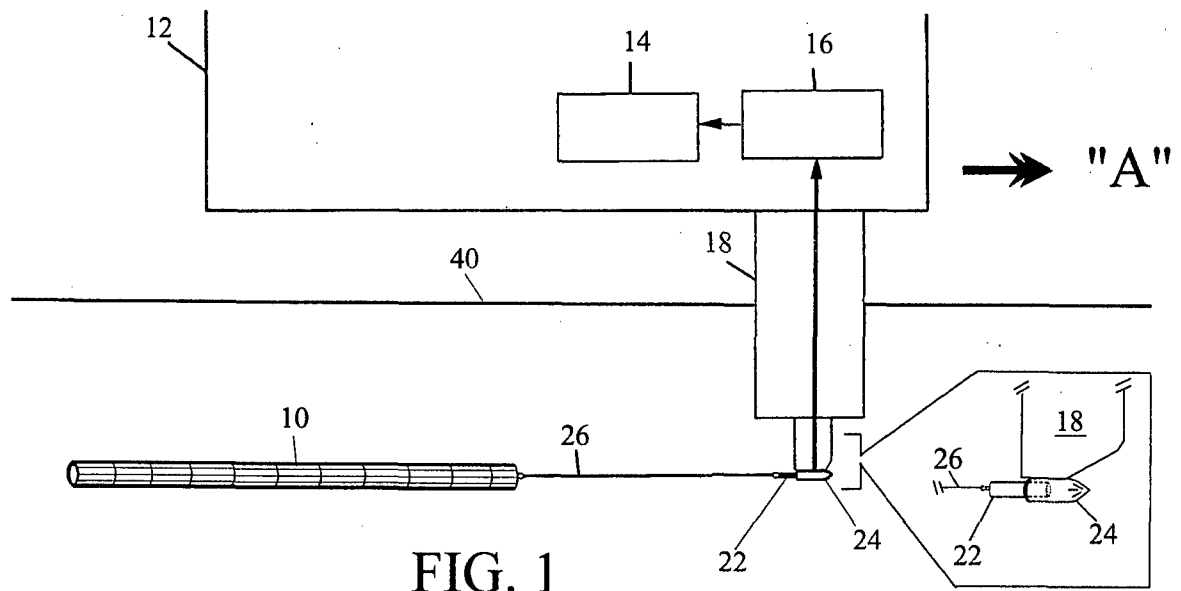


FIG. 1

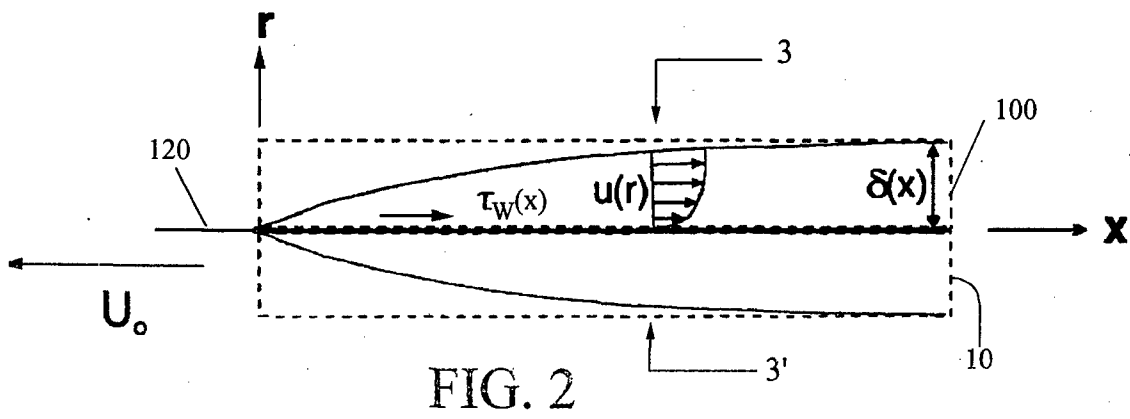


FIG. 2

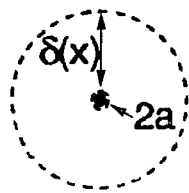


FIG. 3