

DEPARTMENT OF THE NAVY

OFFICE OF COUNSEL NAVAL UNDERSEA WARFARE CENTER DIVISION 1176 HOWELL STREET NEWPORT RI 02841-1708

IN REPLY REFER TO:

Attorney Docket No. 84078 Date: 23 September 2004

The below identified patent application is available for licensing. Requests for information should be addressed to:

PATENT COUNSEL NAVAL UNDERSEA WARFARE CENTER 1176 HOWELL ST. CODE 00OC, BLDG. 112T NEWPORT, RI 02841

Serial Number <u>10/794541</u>

Filing Date 3/1/04

Inventor Peter Hardro

If you have any questions please contact James M. Kasischke, Deputy Counsel, at 401-832-4736.

DISTRIBUTION STATEMENT A Approved for Public Release Distribution Unlimited

Attorney Docket No. 84078 Customer No. 23523

UMBILICAL RETRACTION ASSEMBLY AND METHOD

TO WHOM IT MAY CONCERN:

BE IT KNOWN THAT PETER HARDRO, employee of the United States Government, citizen of the United States of America, resident of Seekonk, County of Bristol, Commonwealth of Massachusetts, has invented certain new and useful improvements entitled as set forth above of which the following is a specification:

JAMES M. KASISCHKE, ESQ. Reg. No. 36562 Naval Undersea Warfare Center Newport, RI Division Newport 02841-1708 TEL: 401-832-4736 FAX: 401-832-1231

1	Attorney Docket No. 84078
2	
3	UMBILICAL RETRACTION ASSEMBLY AND METHOD
4	
5	STATEMENT OF GOVERNMENT INTEREST
6	The invention described herein may be manufactured and used
7	by or for the Government of the United States of America for
8	governmental purposes without the payment of any royalties
9	thereon or therefore.
10	
11	BACKGROUND OF THE INVENTION
12	(1) Field of the Invention
13	The present invention relates generally to remote retraction
14	of umbilical cables which may be utilized in torpedo systems and,
15	more particularly, to assemblies and methods for actively and
16	remotely removing an umbilical cable from a torpedo, prior to
17	launching the weapon from the launch tube.
18	(2) Description of the Prior Art
19	Electrical umbilical cables are utilized to provide
20	communications, power, and other functions to a torpedo in a
21	launch tube. The electrical umbilical cable must be detached
22	prior to launch of a torpedo. The United States Navy has
23	previously launched torpedoes from surface ships via high-
24	pressure air. However, it has been found desirable to provide an
25	advanced surface launcher that eliminates the high-pressure air
	2

· ·

and instead utilizes readily available and highly reliable 1 automotive air bag inflators as the energy source for launching 2 In high-pressure air launching systems, a portion of torpedoes. 3 the high-pressure air energy was used to mechanically detach the 4 electrical umbilical from the weapon just prior to launch. With 5 the elimination of the high-pressure air from the system, the 6 prior art methods of umbilical cord retraction are no longer . 7 functional. 8

9 Previous efforts to solve problems related to the above are10 described by the following patents:

U.S. Patent No. 4,555,130, issued November 26, 1985, to D.W. 11 McClain, discloses a quick-disconnect device for coupling 12 segments of a diver's umbilical hoses includes separable 13 rectangular bodies carrying complementary hose couplings spaced 14 15 therealong. The bodies, which have nested rim and reduced portions, are biased toward separation by a rectangular, flat-16 type spring and releasable retained against separation by a 17 retractable retaining pin extending diagonally therethrough and 18 19 normally locked against retraction by partial rotation.

U.S. Patent No. 4,086,776, issued April 11, 1977, to R. M.
Beard, discloses a lock means for one end of a tension line for
securing the tension line to a subsea installation and
particularly for securing one end of an umbilical line carrying
hydraulic and electrical control circuits to a subsea
installation. The lock means is co-operable with a guide and

locking cylindrical member secured to the subsea installation, 1 2 the cylindrical member having a lock dog receiving recess, and a cylindrical locking device receivable within the cylindrical 3 member and secured to the end of the tension line. The locking 4 device includes a cylindrical housing having a lock dog port and 5 a cylindrical chamber, an elongated annular or sleeve-like piston 6 member within the housing and longitudinally slidable relative to 7. 8 the tension line. The piston member includes a cam surface to 9 slidably engage lock dogs for lateral movement thereof through 10 the lock dog port into locking engagement with the lock recess, a 11 piston portion co-operable with a cylindrical chamber for 12 unlocking the device under fluid pressure, and a piston extension 13 having an upper end exposed through a window for mechanically 14 unlocking the locking device. The piston member is normally 15 biased into locked position by spring means, the locking device 16 being normally non-releasing.

17 U.S. Patent No. 4,258,504, issued March 31, 1981, to H. H. 18 Hicks, discloses a cable release and latch for a security quard 19 hinged to a window with a spring loaded plunger swedged into an 20 outer frame thereof. An armored cable with a flexible inner 21 cable is connected to the spring loaded plunger and extends to a 22 foot treadle operator and shield which may be mounted on the 23 floor inside the opening. Sufficient length of cable is provided 24 to allow free movement of the guard about its hinges. The security guard may thereby be quickly and easily opened without a 25

key by stepping on the foot treadle operator. If the armored
 cable is severed, the spring loaded plunger advances further into
 the keeper which retracts the flexible cable within the armored
 housing to prevent the manual operation of the latch without
 further dismantling of the cable.

U.S. Patent No. 4,300, 254, issued November 17, 1981, to M. 6 7 J. Prior, discloses a release mechanism for use with an 8 underwater exploration device such as one used to release ballast 9 from an ocean bottom seismograph which includes primary and 10 secondary release actuating devices. Ballast weights are secured 11 to the seismograph by cables which themselves are secured by 12 levers held immobile by a removable pin. The pin can be removed 13 by the primary release actuating device, a rotary solenoid with 14 linkage arms which produces a linear stroke, or by the secondary 15 release actuating device, a coil spring held in compression by means of an electroplatable wire around and under the head of the 16 17 pin, or both, allowing the levers to pivot releasing the cables 18 and ballast weight. The mechanical advantage of the levers is 19 utilized to allow a relatively heavy ballast weight to be held 20 while requiring a much smaller force to remove the pin allowing its release. 21

The above described systems do not provide an umbilical release mechanism suitable for use in retracting the cable prior to launching a torpedo including means for avoiding any possibility of entanglement with the fins and propellers.

Consequently, those skilled in the art will appreciate the
 present invention that addresses the above and other problems.

3

4

SUMMARY OF THE INVENTION

5 An object of the present invention is to provide a means to 6 actively and remotely remove an umbilical cable from a vehicle, 7 prior to launching the vehicle from a launch tube.

8 Another object of the present invention is to position the 9 removed umbilical cord within the launch tube such that the fins, 10 shroud, and propeller of the launched vehicle do not become 11 entangled with the umbilical cable.

These and other objects, features, and advantages of the 12 13 present invention will become apparent from the drawings, the 14 descriptions given herein, and the appended claims. However, it 15 will be understood that above listed objects and advantages of 16 the invention are intended only as an aid in understanding 17 various aspects of embodiments of the invention, are not intended 18 to limit the invention in any way, and do not form a 19 comprehensive list of objects, features, and advantages.

The present invention provides an umbilical cord retractor for use in retracting the umbilical cord from a vehicle within a launch tube prior to launch. The umbilical cord retractor comprises one or more elements such as, for instance, a guide arm mountable within the launch tube adjacent the vehicle, a slide mounted within the guide arm and slidable along a predetermined

path formed by the guide arm, a biasing member attached to the 1 2 slide for creating a biasing force to pull the slide away from the vehicle, and/or a locking member to hold the slide in 3 position against a biasing force until the locking member is 4 5 released. Other elements may comprise a linking member attached between the slide and the umbilical cord and/or an activator for 6 releasing the locking member from the slide whereby the slide is 7 pulled by the biasing member and the attached linking member then 8 9 pulls the umbilical cord to a predetermined position with respect 10 to the guide arm. The slide, linking member, biasing member, 11 and/or other components, could be fabricated in one-piece construction for faster assembly, if desired. 12

13 The activator may comprise a solenoid in one embodiment. 14 Likewise, in one embodiment, the locking member may comprise a 15 pin insertable into a matching notch, hole, or groove in the 16 slide. The pin may be angularly oriented with respect to the 17 slide.

18 The biasing member may comprise any type of spring including 19 metal, elastic, or other types of springs.

In operation, a method for retracting an umbilical cord from a vehicle within a launch tube prior to launching the vehicle may comprise one or more steps such as, for instance, attaching a slidable member to the umbilical cord by means of a linking member, biasing the slidable member, locking the slidable member in position within a guide arm, and/or releasing the slidable

member to slide within the guide arm along a predetermined path
 whereby the linking member retracts the umbilical cord and pulls
 it to a predetermined position with respect to the guide arm.

In one embodiment, the step of locking may further comprise inserting a locking pin into the slidable member. The step of releasing may further comprise electrically activating a solenoid. The method may further comprise connecting the solenoid to a push-pull cable and connecting the locking pin to the pushpull cable.

10

11

BRIEF DESCRIPTION OF THE DRAWINGS

12 A more complete understanding of the invention and many of 13 the attendant advantages thereto will be readily appreciated as 14 the same becomes better understood by reference to the following 15 detailed description when considered in conjunction with the 16 accompanying drawing, wherein like reference numerals refer to 17 like parts and wherein:

FIG. 1 is an elevational view showing a retraction assembly connected to a torpedo umbilical cord armed and connected to a torpedo in accord with the present invention; and FIG. 2 is an elevational view showing the torpedo umbilical

22 cord retracted prior to launching the torpedo utilizing the 23 retraction assembly in accord with the present invention.

DESCRIPTION OF THE PREFERRED EMBODIMENT

2	Referring now to the drawings and, more particularly, to
3	FIG. 1 there is shown mechanical umbilical retraction assembly 10
4	attached to an umbilical cable and vehicle in the armed state.
5	The present invention provides a novel mechanical umbilical
6	retraction mechanism 10 which fits within predefined launch tube
7	envelope 11, is actuated remotely, and actively removes the
8	umbilical 30 from vehicle 32, and positions umbilical 30 to the
9	side of the launch tube 11 out of the way of fins 34 and shroud
10	36 of exiting vehicle 32.

11 In the armed state shown in FIG. 1, solenoid 12 is in an extended, de-energized state. Solenoid 12 is connected to push-12 13 pull cable 14 that is routed to an attachment point on the extension or guide arm 16 by suitable means as discussed 14 15 hereinafter. The extension or guide arm 16 and solenoid 12 are fastened to a fixed surface 18 located in launch tube 11. 16 In one 17 presently preferred embodiment, the opposite end of the push-pull cable 14 extends though guide 15 for connection with extension 18 19 arm 16 through securing pin 20. Securing pin 20 is routed through laterally protruding flange 21 which extends outwardly 20 21 from extension arm 16 and is preferably angularly inserted into a 22 groove 23 formed in a slide 22. Securing pin 20 is preferably affixed in laterally protruding flange 21 by push-pull cable 14 23 and extends into slide 22 to thereby prevent slide 22 from moving 24 axially within arm 16 in either direction. Push-pull cable 14 25

could comprise a suitable mechanical linkage of any type. Slide 1 2 22 incorporates features to which a spring 24 and lanyard 26 can 3 be attached. Slide 22 is preferably located inside preferably 4 cylindrical extension arm 16, which may be tubular and/or have 5 lateral guide walls, such that slide 22 can move along a predetermined pathway which is preferably axial but could be 6 7 shaped as desired. Spring 24 is secured at one end to end 25 of 8 the extension arm 16 and on the other end to slide 22. In the 9 armed state, spring 24 is preferably fully extended or extended 10 sufficiently to provide a suitably strong retracting force for pulling on the slide 22 to retract umbilical cable 30, as 11 12 discussed subsequently. The opposite side of the slide 22 is connected to a lanyard 26, which in turn is connected to the 13 14 umbilical cable 30. The umbilical cable 30 is attached to the 15 torpedo 32.

16 FIG. 2 depicts how the mechanical umbilical retraction 17 assembly 10 remotely and actively removes umbilical cable 30 18 prior to torpedo launch. The solenoid 12 is energized thus 19 pulling push-pull cable 14 that in turn pulls securing pin 20 out 20 of slide 22. Once securing pin 20 is removed, then slide 22 is 21 forced down the extension arm 16 in an axial direction away from 22 torpedo 32 by the spring 24. As slide 22 moves axially away from 23 torpedo 32, lanyard 26 is also pulled away from torpedo 32 thus

removing umbilical cable 30 from the torpedo 32 and positioning
 umbilical cable 30 in a predefined position 38 at the end of
 guide arm 16.

It is noted that retraction assembly 10 pulls umbilical 4 cable 30 away from shroud 36 and fins 34, which may be 5 significantly larger than shown in FIG. 2, to thereby prevent any б possibility of entanglement therewith. Moreover, shroud 36 could 7 be located much closer to mount 18, so long as extension arm 16 8 were accordingly adjusted in length to reach to wherever 9 umbilical cable 30 is located on torpedo 32. If necessary, 10 extension or guide arm 16 could extend between vertical and 11 horizontal fins and still remove umbilical cable 30 from the 12 possibility of entanglement due to the ability to provide a 13 predefined position 38 at which umbilical cable 30 will be 14 located after retraction. 15

Accordingly, the present invention provides for the remote 16 and active removal of umbilical cable 30 from torpedo 32 17 positioned within torpedo tube 11. Various alternative designs 18 may be incorporated into the invention. For instance, slide 22 19 may be locked by other means and/or push-pull cable may be 20 activated in other ways. For example, a ball type lock may be 21 used to replace the securing pin 20. Likewise, solenoid could be 22 23 positioned closer to slide and directly in contact with securing Additionally, a linear actuator or rotary solenoid may 24 pin 20. 25 be used to actuate the push-pull cable.

1 Therefore, it will be understood that many additional 2 changes in the details, materials, steps and arrangement of 3 parts, which have been herein described and illustrated in order 4 to explain the nature of the invention, may be made by those 5 skilled in the art within the principle and scope of the 6 invention as expressed in the appended claims.

Attorney Docket No. 84078

1

2

3

4

5

UMBILICAL RETRACTION ASSEMBLY AND METHOD

ABSTRACT OF THE DISCLOSURE

An umbilical cord retractor assembly and method is provided 6 for retracting an umbilical cord from a torpedo or other vehicle 7 prior to launching the vehicle. A guide arm is utilized to guide 8 a slide member along a predetermined path within the guide arm. 9 The slide is connected through a link to the umbilical cord such 10 that the umbilical cord is pulled to a predetermined position 11 within the launch tube out of the way of the fins, shroud, and 12 13 propeller of the vehicle. Prior to activation of the retractor assembly, the slide is affixed in position within the guide arm 14 by a lock. A spring under tension applies a force to the slide. 15 A release cord pulls a pin from the slide to release the cord at 16 17 a desired time, thereby retracting the umbilical cord to the predetermined position within the launch tube. 18



FIG. 1

--

· ·



FIG. 2