



DEPARTMENT OF THE NAVY  
OFFICE OF COUNSEL  
NAVAL UNDERSEA WARFARE CENTER DIVISION  
1176 HOWELL STREET  
NEWPORT RI 02841-1708

IN REPLY REFER TO:

Attorney Docket No. 82646  
Date: 15 March 2002

The below identified patent application is available for licensing. Requests for information should be addressed to:

PATENT COUNSEL  
NAVAL UNDERSEA WARFARE CENTER  
1176 HOWELL ST.  
CODE 00OC, BLDG. 112T  
NEWPORT, RI 02841

Serial Number      09/977,907  
Filing Date        10 October 2001  
Inventor            David A. Tonn

If you have any questions please contact Michael J. McGowan, Patent Counsel, at 401-832-4736.

DIGITAL ANTENNA GONIOMETER AND METHOD

TO ALL WHOM IT MAY CONCERN:

BE IT KNOWN THAT DAVID A. TONN, employee of the United States Government, citizen of the United States of America, and resident of Charlestown, County of Washington, State of Rhode Island, has invented certain new and useful improvements entitled as set forth above of which the following is a specification.

JAMES M. KASISCHKE, ESQ.  
Reg. No. 36562  
Naval Undersea Warfare Center  
Division Newport  
Newport, RI 02841-1708  
TEL: 401-832-4736  
FAX: 401-832-1231

**DISTRIBUTION STATEMENT A**  
Approved for Public Release  
Distribution Unlimited

20020320 058



23523

PATENT TRADEMARK OFFICE

DIGITAL ANTENNA GONIOMETER AND METHOD

TO ALL WHOM IT MAY CONCERN:

BE IT KNOWN THAT DAVID A. TONN, employee of the United States Government, citizen of the United States of America, and resident of Charlestown, County of Washington, State of Rhode Island, has invented certain new and useful improvements entitled as set forth above of which the following is a specification.

JAMES M. KASISCHKE, ESQ.  
Reg. No. 36562  
Naval Undersea Warfare Center  
Division Newport  
Newport, RI 02841-1708  
TEL: 401-832-4736  
FAX: 401-832-1231

1 Attorney Docket No. 82646

2

3 DIGITAL ANTENNA GONIOMETER AND METHOD

4

5 STATEMENT OF THE GOVERNMENT INTEREST

6 The invention described herein may be manufactured and used  
7 by or for the Government of the United States of America for  
8 Governmental purposes without the payment of any royalties  
9 thereon or therefore.

10

11 CROSS-REFERENCE TO OTHER PATENT APPLICATIONS

12 Not applicable.

13

14 BACKGROUND OF THE INVENTION

15 (1) Field of the Invention

16 The present invention relates to antennas and, more  
17 particularly, to a digital goniometer especially suitable for  
18 controlling a steerable antenna pattern from two orthogonal  
19 antennas over the VLF/LF communications band.

20 (2) Description of the Prior Art

21 Submarine antenna systems may frequently utilize the VLF/LF  
22 communications band (8kHz - 200 kHz) by means of two magnetic  
23 loop antennas which may be referred to as a multifunction

1 antenna. The two magnetic loop antennas are preferably  
2 identical and aligned perpendicular to each other in one  
3 housing. One loop is referred to as the Fore and Aft (F/A)  
4 antenna. The other loop is referred to as the Athwartships  
5 (ATH) antenna. The multifunction antenna does not rotate, and  
6 there are null spots where the antenna gain is near zero.  
7 Orienting this null spot toward a source of interference is  
8 desirable to reduce the received interference. Likewise,  
9 orienting the highest gain region of the antenna toward the  
10 desired signal is also desirable. Currently, an analog  
11 mechanical device is used to combine the F/A signal with the ATH  
12 signal in order to obtain the desired orientation. The  
13 mechanical device contains a stator for each loop and a rotor  
14 that can be rotated to adjust the proportional contribution of  
15 each loop. The device employs Faraday's Law of Induction for  
16 its operation. The mechanical device presently utilized is an  
17 electromechanical resolver often known as a goniometer. While  
18 goniometers may be utilized to achieve other effects and  
19 functions, a goniometer may also be utilized for steering a  
20 figure-eight antenna pattern with two identical orthogonal  
21 antennas by mechanical means.

1       It would be desirable to provide a more accurate,  
2   lightweight, compact, antenna goniometer that does not require  
3   elaborate external drive circuitry and has no moving parts.

4       Patents that show attempts to solve the above and other  
5   related problems are as follows:

6       U.S. Patent No. 5,402,132, issued March 28, 1995, to Hall  
7   et al., discloses a direction finding system utilizing a single  
8   monopole/crossed slot antenna in combination with associated  
9   electronic circuitry. Each of the four ports of the antenna are  
10   connected to amplitude varying elements whose outputs are  
11   combined by a power combiner. A microcontroller stores the  
12   signal. The microcontroller also generates the signals which  
13   control the amplitude variation supplied by each of the  
14   amplitude varying elements. Two configurations of the amplitude  
15   varying elements are provided by predetermined settings within  
16   the microcontroller. The two settings are selected to configure  
17   the antenna on two distinct reception patterns, each with a  
18   predetermined angular offset from a reference direction. In  
19   operation the antenna will be configured for a first reception  
20   pattern. The signals received will be detected, quantified, and  
21   stored by the microcontroller which subsequently reconfigures  
22   the antenna for a second reception pattern. A second signal is  
23   received by the antenna in the second configuration and again is

1 detected, quantified, and stored by the microcontroller. The  
2 microcontroller determines the difference between the two  
3 signals and, based upon a predetermined calibration curve  
4 relating differences in the stored signals to the angle of  
5 arrival of the signals from the reference direction, determines  
6 the angle from which the signal arrived.

7 U.S. Patent No. 6,160,519, issued December 12, 2000, to  
8 Christian O. Hemmi, discloses a two-dimensionally steered  
9 antenna system including a planar lensing system operable to  
10 focus signals received from a plurality of ground-based cells. A  
11 first steering system is operable to steer a beam for each  
12 ground-based cell in a first direction by weighing signals  
13 associated with the ground-based cell based on a position of the  
14 antenna system relative to the ground-based cell in the first  
15 direction. A second steering system is operable to steer the  
16 beam for each ground-based cell in a second direction by  
17 weighing signals associated with the ground-based cell based on  
18 a position of the antenna system relative to the ground-based  
19 cell in the second direction.

20 U.S. Patent No. 4,150,382, issued April 17, 1979, to Ray J.  
21 King, discloses an invention which provides a guided wave  
22 antenna having a radiation pattern which can be controlled

1 electronically, by control signals derived from a computer or  
2 any other suitable source. In this way, the directional  
3 characteristics of the antenna can be adjusted and/or scanned  
4 rapidly, without any mechanical manipulation of the antenna. In  
5 one embodiment, a guided radio wave is launched along an antenna  
6 surface having an array of elements which provide variable non-  
7 uniform surface impedance adapted to be controlled by electronic  
8 signals. For example, each variable impedance element may  
9 comprise a wave guide section having one end leading from the  
10 antenna surface. Each wave guide section may include a solid-  
11 state electronic reflection amplifier having characteristics  
12 which can be varied by supplying control signals to the  
13 amplifier, to vary the magnitude and phase angle of the wave  
14 reflected from the reflection amplifier. By changing the control  
15 signals supplied to any particular reflection amplifier, it is  
16 possible to cause attenuation or amplification and phase shift  
17 of the guided wave as it passes across the particular wave guide  
18 section. A wide variety of solid-state electronic control  
19 elements may be provided along one or more surfaces of the  
20 antenna. In another embodiment, a wave traveling in a closed  
21 subsurface wave guide is coupled into the guided wave open  
22 surface structure, using an array of wave guide elements

1 containing electronically controllable amplifiers and phase  
2 shifters.

3 U.S. Patent No. 5,714,961, issued February 3, 1998, to Kot  
4 et al., discloses a directional planar antenna. The antenna has  
5 an array of coaxial ring-slot radiating elements formed through  
6 a conductive layer on a dielectric substrate. A number of  
7 probes, coupled to the ring-slot elements, selectively excite a  
8 separate resonant mode on each ring-slot element. The resonant  
9 mode supported by a ring-slot element depends upon the geometry  
10 of that ring-slot element. The resonant modes combine in the far  
11 field to form a radiation pattern directional in azimuth and  
12 elevation. By adjustment of the relative phase difference or  
13 relative amplitude between the excited modes, the radiation  
14 pattern can be steered.

15 The above-cited prior art does not show a suitable  
16 lightweight, compact antenna goniometer with non-moving  
17 components utilizing two orthogonal antennas that is operable  
18 for steering a figure-eight antenna pattern. Those skilled in  
19 the art will appreciate the present invention that addresses the  
20 above and other problems.

1 SUMMARY OF THE INVENTION

2 Accordingly, it is an object of the present invention to  
3 provide an improved antenna goniometer.

4 It is another object of the present invention to provide an  
5 antenna goniometer operable with two orthogonal antennas for  
6 steering a figure-eight antenna pattern.

7 An advantage of a system in accord with the present  
8 invention is an antenna goniometer that can be operated in a  
9 fully automated manner that can be controlled from software.

10 In accordance with the present invention, a VLF/LF  
11 communications antenna goniometer for a first antenna loop and a  
12 second antenna loop is provided that is operable for controlling  
13 an angular orientation of a figure-eight antenna reception  
14 pattern for the first antenna loop and the second antenna loop.  
15 The preferred embodiment shows first and second antennas which  
16 are orthogonal with respect to each other; however, the  
17 teachings of this invention can be applied by one of ordinary  
18 skill in the art to other antenna configurations. The  
19 goniometer comprises one or more elements such as, for instance,  
20 a first variable gain amplifier for a first signal produced  
21 utilizing the first antenna wherein the first variable gain  
22 amplifier has a first output, a first data register for storing  
23 a digital value related to a first gain setting for the first

1 variable gain amplifier such that the first gain setting is  
2 related to the angular orientation, a second variable gain  
3 amplifier for the second antenna for a second signal produced  
4 utilizing the second antenna wherein the second variable gain  
5 amplifier has a second output, a second data register for  
6 storing a digital value related to a second gain setting for the  
7 second variable gain amplifier such that the second gain setting  
8 is related to the angular orientation, and a summing amplifier  
9 for combining the first output and the second output.

10 Other elements of the goniometer may comprise a first  
11 digital data link for supplying the first gain setting to the  
12 first data register and a second digital data link for supplying  
13 the second gain setting to the second data register. In the  
14 preferred embodiment, the first gain setting is a sine function  
15 related to the angular orientation, and the second gain setting  
16 is a cosine function related to the angular orientation.

17 The goniometer may further comprise an inverter for  
18 inverting at least one of the first signal or the second signal.  
19 The summing amplifier adds the first output and the second  
20 output together.

21 In operation, a method is provided for controlling an  
22 antenna reception pattern for the first antenna and second  
23 antenna wherein the method comprises one or more steps such as,

1 for instance, producing a first antenna signal utilizing the  
2 first antenna, producing a second antenna signal utilizing the  
3 second antenna, determining a first gain signal related to a  
4 desired antenna reception pattern, determining a second gain  
5 setting related to the desired antenna reception pattern,  
6 digitally adjusting the first gain setting for the first antenna  
7 signal to produce a first antenna processed signal, digitally  
8 adjusting the second gain setting for the second antenna signal  
9 to produce a second antenna processed signal, and combining the  
10 first antenna processed signal and the second antenna processed  
11 signal. Additional steps may include determining the first gain  
12 signal and the second gain signal to produce a figure-eight  
13 reception pattern having a desired angular orientation.

14 Other method steps may include providing that the first  
15 antenna signal and the second antenna signal are substantially  
16 in phase with each other to produce a figure-eight pattern.  
17 Additional steps may include utilizing software to adjust the  
18 first gain setting and the second gain setting.

19 In more detail, the method may comprise controlling a first  
20 magnitude of the first antenna signal with the first gain  
21 setting to produce the first antenna processed signal, and  
22 controlling a second magnitude of the second antenna signal with  
23 the second gain setting to produce the second antenna processed

1 signal and/or providing that the first gain setting is related  
2 to a sine function of an angular orientation of a figure-eight  
3 reception pattern, and providing the second gain setting is  
4 related to a cosine function of the angular orientation of the  
5 figure-eight reception pattern.

6 In other words, a method is provided to produce a figure-  
7 eight antenna pattern with a selected angular orientation  
8 comprising one or more steps such as, for instance, providing  
9 that a first antenna signal and a second antenna signal are  
10 substantially in phase with respect to each other, and digitally  
11 controlling a first magnitude of the first antenna signal and a  
12 second magnitude of the second antenna signal such that the  
13 first magnitude is related to a sine function of the angular  
14 orientation and the second magnitude is related to cosine  
15 function of the angular orientation. Other method steps then  
16 include combining the first antenna signal with the second  
17 antenna signal. The combining may further comprise summing the  
18 first antenna signal to the second antenna signal. Preferably  
19 the method comprises utilizing software to compute a first gain  
20 setting for the first magnitude and a second gain setting for  
21 the second magnitude, storing the first gain setting in a first  
22 register and storing the second gain setting in a second

1 register. Preferably the method permits for dynamically  
2 changing the angular orientation utilizing software.

3       These and other objects, features, and advantages of the  
4 present invention will become apparent from the drawings, the  
5 descriptions given herein, and the appended claims. It will be  
6 understood that above listed objects and advantages of the  
7 invention are intended only as an aid in understanding aspects  
8 of the invention, are not intended to limit the invention in any  
9 way, and do not form a comprehensive list of objects, features,  
10 and advantages.

11

#### 12                   BRIEF DESCRIPTION OF THE DRAWING

13       A more complete understanding of the invention and many of  
14 the attendant advantages thereto will be readily appreciated as  
15 the same becomes better understood by reference to the following  
16 detailed description when considered in conjunction with the  
17 accompanying drawing wherein corresponding reference characters  
18 indicate corresponding parts and wherein:

19       FIG. 1 is a schematic diagram showing a pair of antennas  
20 perpendicular to each other in a coordinate system;

21       FIG. 2 is a schematic diagram of a digital antenna  
22 goniometer in accord with the present invention; and

1        FIG. 3 is a block diagram of a goniometer operable for  
2 steering a VLF/LF figure-eight antenna pattern.

3  
4        BRIEF DESCRIPTION OF THE PREFERRED EMBODIMENTS

5        Referring now to the drawings, and more particularly to  
6 FIG. 1, there is shown a circuit diagram of antenna goniometer  
7 10 in accord with the present invention. Antenna goniometer 10  
8 provides a more lightweight, compact antenna goniometer which  
9 does not require elaborate external drive circuitry.  
10 Furthermore, antenna goniometer 10 has no moving parts and is  
11 fully controllable by software from a bus such as a TTL parallel  
12 bus.

13        FIG. 2 discloses antenna system 12 with electrically short  
14 electric loop antennas 14 and 16 perpendicular to each other and  
15 mapped onto a coordinate system which utilizes spherical  
16 coordinates as indicated, such that antenna 16 is aligned with  
17 the z-axis and antenna 14 is aligned with the y-axis. This  
18 arrangement represents two crossed electric loops, but because  
19 of duality, it can also be used to represent a pair of  
20 orthogonal magnetic loop antennas.

21        Beginning from principles of the patterns of individual  
22 electrically short loops, equations governing the tandem  
23 performance can be derived and the conditions under which

1 steerable figure-eight responses can be determined. Thus, it  
2 can be shown that the square of the magnitude of the field  
3 pattern of antenna system 12 is:

4  
5 
$$|H_{tot}|^2 = K \cdot |\sin \theta \cos \alpha - \cos \theta + j \sin \theta \sin \alpha|^2 = K(1 - \sin 2\theta \cos \alpha), \quad (1)$$

6  
7 where:

8  $\Phi$  is replaced with  $90^\circ$  because we are concerned with field  
9 patterns in the y-z plane,

10  $\alpha$  is the phase difference of the current feeds for the  
11 antennas wherein the feed currents are of equal magnitude, and  
12 K is a constant.

13 In the event that  $\alpha = 0^\circ$  (i.e., the two antennas are fed in  
14 phase), the resulting pattern is a figure eight, rotated so that  
15 its null axis is aligned with  $45^\circ$  on the polar scale. However,  
16 as  $\alpha$  is increased (the antennas now being fed out of phase), the  
17 null begins to vanish, leaving a peanut-shaped pattern. It can  
18 also be shown that the net power varies only by approximately  
19 1.5 dB over its domain for an  $80^\circ$  phase shift. The exact amount  
20 of tolerance allowed on  $\alpha$ , then, would depend on the amount of  
21 loss that can be tolerated due to the depth of the null in the

1 pattern. If high loss can be tolerated, a looser tolerance on  $\alpha$   
2 is allowed.

3 For figure-eight pattern generation, consider now the more  
4 general case of two arbitrary antenna feed currents with an  
5 arbitrary phase difference between them. Suppose the currents  
6 are:

$$7 \quad I_z = Ae^{j\alpha}, \quad I_y = B. \quad (2)$$

8 It can then be shown that the resulting total magnetic  
9 field produced by this excitation will be

10

$$11 \quad \vec{H}_{tot} = j\beta \Delta L \frac{e^{-j\beta r}}{4\pi r} (Ae^{j\alpha} \sin\theta - B \cos\theta) \hat{\phi}, \quad (3)$$

12

13 where:

14  $\Delta L$  is the length of the antennas, and

15  $\beta$  is the wavenumber in the medium that the fields propagate  
16 through.

17 To determine the farfield power pattern, the magnitude must  
18 be computed:

19

$$20 \quad |H_{tot}|^2 = \left| j\beta \Delta L \frac{e^{-j\beta r}}{4\pi r} (Ae^{j\alpha} \sin\theta - B \cos\theta) \right|^2 = K \cdot |A(\cos\alpha + j \sin\alpha) \sin\theta - B \cos\theta|^2. \quad (4)$$

1        Again the term K represents a term that is not dependent on  
2        the polar coordinate  $\theta$  and does not need to be considered in  
3        determining the shape of the power pattern. Performing some  
4        simplifications, we find that this expression reduces to

5

6         $|H_{tot}|^2 = A^2 \sin^2 \theta + B^2 \cos^2 \theta - AB \cos \alpha \sin 2\theta.$  (5)

7

8        To determine the possible ways of using the above equation  
9        to give steerable figure eight.

10       If we make the substitution  $X = A \sin \theta$ ,  $Y = B \cos \theta$ , the  
11       power pattern reduces to

12

13        $|H_{tot}|^2 = X^2 + Y^2 - 2XY \cos \alpha.$  (6)

14

15       This is a form recognizable as the Law of Cosines from  
16       trigonometry. Using this knowledge, we can draw some  
17       conclusions about the possible existence of nulls in the  
18       pattern. If desired, we can visualize the power pattern's  
19       relationship to X and Y as a triangle where X and Y are adjacent  
20       legs with angle  $\alpha$  between the legs and  $|H_{tot}|^2$  is the leg opposite  
21       angle  $\alpha$ . Clearly, if  $\alpha = 90^\circ$ , then the triangle is a right  
22       triangle, with the magnitude of the power patter as its

1 hypotenuse. Such a triangle cannot, according to the  
 2 Pythagorean Theorem, have a zero length hypotenuse unless both  
 3 legs of the triangle are of zero length. Therefore, it is not  
 4 possible to generate a pattern containing nulls at all with a  
 5 90 phase shift between the antennas. It is also apparent that  
 6 for angles  $\alpha \neq 0^\circ$ , it is not possible for a null to occur unless,  
 7 of course, both X and Y are zero. This, however, is an  
 8 impossibility, since X depends on  $\sin \theta$ , and Y depends on  $\cos \theta$ ,  
 9 and these functions are never zero simultaneously.

10 To form a true null in the pattern, then, it is necessary  
 11 that  $\alpha = 0^\circ$  (i.e., the two signals must be in phase). For in-  
 12 phase currents, then the power pattern is given by

$$14 \quad |H_{tot}|^2 = A^2 \sin^2 \theta + B^2 \cos^2 \theta - 2AB \sin \theta \cos \theta = (A \sin \theta - B \cos \theta)^2. \quad (7)$$

16 To locate the nulls in the pattern, we solve for angles  $\theta$   
 17 that render the equation zero:

$$19 \quad A \sin \theta - B \cos \theta = 0 \Rightarrow \tan \theta = \frac{B}{A} \Rightarrow \theta = \arctan\left(\frac{B}{A}\right) (-90 \leq \theta \leq +90). \quad (8)$$

20 We see here, then, that a single null axis exists for the  
 21 pattern and that its position is wholly dependent on the

1 magnitudes of the currents feeding the antennas. By appropriate  
2 choices of A and B, then, we can steer the null in the pattern  
3 to any desired location. From the point of view of  
4 implementation, it would be desirable to remove the arctangent  
5 from the computation. This is accomplished by choosing A and B  
6 such that  $A = \cos \gamma$ ,  $B = \sin \gamma$ . In this way, the argument of the  
7 arctangent function is the tangent of the angle  $\gamma$ , the angle to  
8 which the null is to be steered. The resulting power pattern  
9 will be given by:

10

11  $|H_{tot}|^2 = (\sin \theta \cos \gamma - \cos \theta \sin \gamma)^2 = \sin^2(\theta - \gamma).$  (9)

12

13 This pattern may be recognized as that of a standard  
14 figure-eight antenna pattern rotated by  $\gamma$  degrees. The null in  
15 the pattern remains well defined for low phase differences; the  
16 exact tolerance on  $\alpha$  will depend on the requirement for the null  
17 depth of the figure-eight pattern.

18 In a real network, it will not always be possible to create  
19 a situation where the two signals from the antennas possess the  
20 exact magnitudes and phases to produce a perfect omni or figure-  
21 eight pattern. It is of value to see what effect the errors in

1 magnitude and phase will have on the resulting patterns for the  
2 figure-eight antenna.

3 The location of the minimum in the pattern can be  
4 determined where the value of  $\theta$  satisfies

$$6 \quad \frac{\partial}{\partial \theta} |H_{tot}|^2 = 0 \Rightarrow \frac{\partial}{\partial \theta} (A^2 \sin^2 \theta + B^2 \cos^2 \theta - AB \cos \alpha \sin 2\theta) = 0. \quad (10)$$

7  
8 By working out the partial derivative, it can be shown that  
9 for positive values of A and B the pattern minimum occurs for

$$11 \quad \theta_{\min} = \frac{1}{2} \arctan \left( \frac{2AB \cos \alpha}{A^2 - B^2} \right) + \frac{m\pi}{2} \quad m = 0, 2, 4, \dots \quad (11)$$

12  
13 It can also be shown that the pattern maximum occurs 90  
14 away from the pattern minimum:

$$16 \quad \theta_{\max} = \frac{1}{2} \arctan \left( \frac{2AB \cos \alpha}{A^2 - B^2} \right) + \frac{n\pi}{2} \quad (n = 1, 3, 5, \dots) = \theta_{\min} + \frac{\pi}{2} \quad (12)$$

17  
18 The min-to-max ratio can now be determined by substituting  
19 the above values of  $\theta$  into the expression for the general power  
20 pattern. It can be shown that the resulting M is given by:

$$M = \frac{A^2 \sin^2 \theta_{\min} + B^2 \cos^2 \theta_{\min} - AB \cos \alpha \sin 2\theta_{\min}}{A^2 \cos^2 \theta_{\min} + B^2 \sin^2 \theta_{\min} + AB \cos \alpha \sin 2\theta_{\min}}. \quad (13)$$

From these equations, it can be seen that errors in the magnitudes and phase of the currents on the antennas can lead to non-ideal performance when the signals are combined to produce the figure-eight pattern. For instance, a non-zero phase difference  $\alpha$  will cause a position error in the location of the null in a figure-eight pattern.

FIG. 3 shows a functional block diagram of system 10A to combine two orthogonal patterns into omni and/or steerable figure-eight patterns. The two antenna inputs 18 and 20 are referred to as the F/A (fore/aft) and ATH (athwart) antennas, respectively. Variable gain amplifiers 22 and 24 have a gain of  $\cos \gamma$  and  $\sin \gamma$ , respectfully, where the angle  $\gamma$  input to steer to is indicated at 26. Summation output 32 produces a steerable figure-eight output.

The details of an electronic circuit to perform the functions shown in FIG. 3 for a steerable figure-eight output are indicated in FIG. 2. This circuit is specifically tailored to VLF/LF radio waves. One of ordinary skill in the art could devise a similar circuit for higher frequencies using monolithic components. In FIG. 2, F/A antenna input is preferably provided

1 by a 50Ω cable to terminals F/A Hi and F/A Lo, designated as 34  
2 and 36, respectively. The F/A antenna input signal is then  
3 isolated by transformer 38. Likewise, the ATH antenna input is  
4 provided to terminals ATH Hi and ATH Lo, designated as 40 and  
5 42, respectively, by a 50Ω cable and is isolated by transformer  
6 44. Capacitors 46 and 48 provide filtering on the input  
7 signals. Inversion of the ATH channel, if needed depending on  
8 the desired quadrant of the antenna pattern, is provided by  
9 relay 50 in response to an invert signal on line 60. Relay 50  
10 selectively reverses the polarity of the ATH signal before the  
11 ATH signal arrives at the ATH variable gain amplifier which is  
12 comprised of digital-to-analog component (DAC) 52 and amplifier  
13 54 which components may preferably be connected together in a  
14 feedback loop to form a variable gain amplifier. The F/A signal  
15 is applied to F/A variable gain amplifier which is comprised of  
16 DAC 56 and amplifier 58 preferably connected together in a  
17 feedback loop to form a variable gain amplifier.

18 The gain settings for the F/A variable gain amplifier and  
19 the ATH variable gain amplifier are supplied by F/A steering  
20 input 62 and ATH steering input 64 by software controlling the  
21 system. Thus, inputs 62 and 64 may typically each comprise a  
22 data bus. The F/A gain setting is stored in a register within  
23 DAC 56 and the ATH gain setting is stored in a register with DAC

1 52. The software for antenna goniometer 10 computes, from a  
2 given null steering angle  $\phi$ , the required gains of the F/A and  
3 ATH channels:  $\sin \phi$  and  $\cos \phi$ , respectively. Inversion of the  
4 ATH signal by inverter 50 is provided to allow steering into the  
5 2nd and 4th quadrants of the plane of the antenna pattern.

6 Trimmer potentiometers 66 and 68 provide zero-offset  
7 adjustment for the variable gain stages. Capacitors 70 and 72  
8 shunt any out-of-band high frequency (HF) interference from the  
9 signal lines. The scaled signals are then fed to summing  
10 amplifier or processor 74 formed by amplifier 76 and resistors  
11 78, 80, and 82. Variable resistor 82 provides a manual gain  
12 adjustment. The summation of signals is provided at junction  
13 84.

14 Buffer amplifier 86, together with resistor 88 and  
15 transformer 90 are preferably utilized to provide a  $50\Omega$  drive  
16 stage to drive the balanced output lines 92 and 94.

17 Thus, the operation of circuit 10 is as follows and is in  
18 accord with the mathematical theory of steerable antenna  
19 patterns described hereinbefore. The signals from the  
20 orthogonal antennas denoted herein by F/A (Fore/Aft) and ATH  
21 (Athwart) enter the system as described hereinbefore at inputs  
22 34, 36 and 40, 42, respectively. The signals are directed to  
23 respective variable gain amplifiers. The F/A variable gain

1 amplifier is comprised of DAC 56 and amplifier 58. The ATH  
2 variable gain amplifier is comprised of DAC 52 and amplifier 54.  
3 The gain settings for the variable gain amplifiers are  
4 preferably stored in a register in the respective DAC 56 and DAC  
5 52. The gain settings come from the controlling software. The  
6 software has a F/A steering link input at 62 and an ATH steering  
7 link at 64. Thus, links 62 and 64 preferably comprise suitable  
8 digital data busses that are used for inputting the gain  
9 settings into the respective registers within DAC 52 and DAC 56.  
10 Software is preferably utilized to compute the required gain  
11 settings from a given null steering angle and input the gain  
12 settings whenever desired. The gain settings may be relatively  
13 constant or may change to create some desired pattern, such as a  
14 rotating figure-eight pattern, with a desired timing such as a  
15 desired number of rotations per minute. An inversion of the ATH  
16 signal is provided by inverter 50 to allow steering into the 2nd  
17 and 4th quadrants of the plane of the antenna pattern. The  
18 resultant signals from the variable gain amplifier are added by  
19 a summing amplifier and presented to the output at 92 and 94.

20 While a particular circuit for a presently preferred  
21 embodiment has been described, it will be understood that the  
22 invention is not limited to this particular circuit and could be  
23 implemented in other ways.

1        It will be appreciated by those skilled in the art that the  
2    invention can also be implemented using a suitable programmed  
3    general purpose computer and/or special purpose hardware, with  
4    program routines or logical circuit sets performing as  
5    processors. Program routines, electronic circuits, or logical  
6    circuit sets may also be referred to as processors or the like.

7        It will be understood that many additional changes in the  
8    details, materials, steps and arrangement of parts, which have  
9    been herein described and illustrated in order to explain the  
10   nature of the invention, may be made by those skilled in the art  
11   within the principle and scope of the invention as expressed in  
12   the appended claims.

3 DIGITAL ANTENNA GONIOMETER AND METHOD

5 ABSTRACT OF THE DISCLOSURE

6 A digital goniometer and steering method are provided which  
7 may be used, in a preferred embodiment, for producing a figure-  
8 eight antenna reception pattern oriented at any selectable angle.  
9 The digital goniometer works with the signals from a pair of  
10 orthogonal antenna loops. In a presently preferred embodiment a  
11 first register and a second register are provided for a respective  
12 first variable gain amplifier and a second variable gain amplifier.  
13 The first and second registers receive from software a digital  
14 value related to a sine function and cosine function of the  
15 selectable angle. The first and second registers are utilised to  
16 set the gain of the first and second variable gain amplifiers. The  
17 outputs of the first and second variable gain amplifiers are added  
18 to produce the reception signal.

FIG. 1

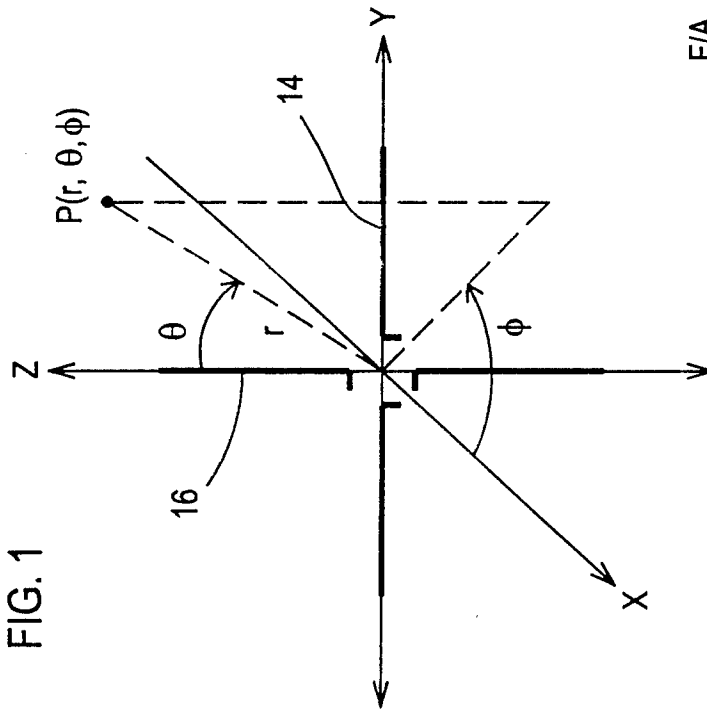


FIG. 3

