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LANDMASS FLY-AROUND BOUNDARY GENERATION

STATEMENT OF GOVERNMENT INTEREST

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BACKGROUND OF THE INVENTION

(1) Field of the Invention

This invention is directed to guidance systems and processes, and more particularly, to a process for generating fly-around boundaries relative landmasses for projectiles such as missiles.

(2) Description of the Prior Art

Automated flight missiles such as the Tomahawk missile typically include guidance systems which are preprogrammed for guiding the missile to its destination. In order to avoid radar, defenses and to maintain rules of engagement, the missiles are programmed to fly at a standoff distance from the coastline of landmasses. With currently available technology, operators of the missiles or guidance systems work to manually calculate and plot fly-around boundary distances from selective landmasses in

1 order to identify missile waypoint positions. This process is
2 time consuming and results in over-estimated fly-around boundary
3 distances which increase the length of the flight path, thus
4 reducing the missile's capability.

5 The prior art includes U.S. Patent No. 5,150,295 directed to
6 a computerized system for joining individual maps into a single
7 map product. It discloses an improved process for making a
8 larger map from individual 7.5 digital line graph (DLG) data.
9 The process is fully automated and performed by computer with
10 minimal human interaction, eliminating errors and producing a
11 more accurate final map product. The process includes conversion
12 of the raw DLG data files into ARC/INFO format, locating the
13 border arcs of each individual data set, edgematching the
14 individual map data sets, and joining the data sets into a
15 single, large map coverage. Any node along the border arc which
16 cannot automatically be edgematched is noted in a special error
17 file. A geographer then matches the unmatched edges which
18 contain an error in the input data. A large map product is
19 provided as the product of the process. However, nothing is
20 disclosed in this patent related to the automated production of
21 fly-around boundaries relative landmasses.

22 There exists a need for a process for automatically
23 computing fly-around boundaries relative to any form of landmass
24 based on cartographic data inputted into the process.

SUMMARY OF THE INVENTION

1
2 The primary object of this invention¹ is to provide an
3 automated process for determining fly-around boundaries relative
4 to any form of landmass for the guidance of projectiles or the
5 like.

6 Another object of this invention is to provide an automated
7 process for creating fly-around boundaries which produces a
8 flight path defined by the boundaries by eliminating small
9 harbors and inlets of coastlines from a model of a landmass
10 coastline.

11 Still another object of this invention is to provide a
12 process for automatically creating fly-around boundaries based on
13 a desired standoff distance from a landmass coastline.

14 Still another object of this invention is to provide a
15 process for creating fly-around boundaries for landmasses, which
16 combines the fly-around boundaries for adjacent intersecting
17 landmass boundaries into a resultant fly-around boundary,
18 depending upon the proximity of the landmass boundaries.

19 The foregoing objects are attained by the process of the
20 present invention for generating fly-around boundaries for use by
21 projectiles or the like, comprising the steps of providing
22 original cartographic data representative of at least one
23 geographical position on a landmass and providing a predetermined
24 value for the spacing of a fly-around boundary from the
25 geographical position; digitizing the original cartographic data

1 and creating a landmass model in a format which includes
2 latitudinal and longitudinal coordinates' of the geographical
3 position; using as an input the digitized data and the
4 predetermined value into a means for analyzing the digitized
5 data; generating new data representative of the fly-around
6 boundary based on the predetermined value and the land mass
7 model, via the analyzing means; and providing a navigational
8 control means for steering the projectile and using as input the
9 new data into the navigational control means for controlling the
10 flight path of the projectile along the boundary.

11 The process of the present invention further includes the
12 filtered landmass model having a landmass side and a water side.

13 The step of generating fly-around boundaries further comprises
14 the step of offsetting original fly-around segments from segments
15 generated from said new data by a distance equal to the
16 predetermined value at an orientation substantially parallel to
17 the segments and toward said water side for setting a fly-around
18 boundary.

19 The process of the present invention further includes the
20 cartographic data being provided for a plurality of landmasses,
21 wherein the means for analyzing carries out the steps of
22 determining the fly-around segments for each of the plurality of
23 landmasses and combining the fly-around segments of each of the
24 plurality of landmasses for creating a resultant fly-around
25 boundary.

1 The details of the present invention are set out in the
2 following description and drawings wherein like reference
3 characters depict like elements.
4

5 BRIEF DESCRIPTION OF THE DRAWINGS

6 FIG. 1 is a schematic view of an operating system in which
7 the process of the present invention can be used.

8 FIGS. 2A and 2B are flow diagrams of one tier of the process
9 performed by the process of the present invention directed to the
10 elimination of small harbors or inlets and the like from a
11 landmass model.

12 FIGS. 3A, 3B and 3C are schematic diagrams of a coastline
13 and illustrate the results of the sequence of steps carried out
14 by the process of FIGS. 2A and 2B for removing small harbors and
15 the like and cross overs from the landmass model.

16 FIGS. 4A, 4B and 4C represent a flow diagram of a second
17 tier of the process carried out by the process of the present
18 invention for creating fly-around boundaries relative the
19 landmass model resulting from the process outlined in FIG. 2A and
20 2B.

21 FIGS. 5A and 5B are schematic diagrams of fly-around
22 boundaries for particular types of landmasses including a single
23 point island and a line segment island.

24 FIG. 6 is a schematic representation of a fly-around
25 boundary created for a landmass having a convex configuration.

1 FIG. 7 is a schematic representation of a fly-around
2 boundary created for a landmass having a concave configuration.

3 FIG. 8 is a schematic representation similar to FIG. 6
4 showing the fly-around boundary created for a convex segment of
5 landmass and also indicating steps in the process of the present
6 invention for correcting standoff error associated with fly-
7 around boundaries for convexly configured landscapes.

8 FIG. 9 is a further schematic view of a convexly configured
9 landmass adjacent a concavely configured landmass and a process
10 for correcting errors associated with the creation of fly-around
11 boundaries for adjacent convex-concave configured landmasses.

12 FIG. 10 is a diagram of a section of a non-island landmass
13 illustrating the results of a process for extending end fly-
14 around segments

15 FIG. 11 is a flow diagram of a third tier of the process
16 used by the present invention for combining fly-around boundaries
17 of landmasses having intersecting boundaries.

18 FIG. 12A, 12B and 12C are diagrams of particular types of
19 landmasses having intersecting fly-around boundaries and
20 illustrate the process for creating combined boundaries.

21 FIG. 13A, 13B, 13C, 13D and 13E are diagrams showing the
22 process for combining fly-around boundaries for a plurality of
23 island type landmasses in accordance with the process of FIG. 10.

1 DETAILED DESCRIPTION OF THE PREFERRED EMBODIMENT

2 FIG. 1 is a schematic diagram of a navigational control or
3 planning system in which the landmass fly-around boundary process
4 of the present invention is utilized. The landmass boundary
5 generation process is implemented with the aid of a computer 10
6 associated with a large navigational control system 13 used to
7 plan the flight path and automatic engagement of a projectile
8 such as a missile

9 The process steps discussed below for FIGS. 2A and 2B, 4A
10 and 4B and FIG. 11 are implemented using computer 10, which may
11 be of any applicable kind in the art. Accordingly, with respect
12 to the present invention, cartographic data and a desired
13 standoff distance S (see FIG. 6) are used as input into a storage
14 or memory device 11 associated with computer 10 for use by the
15 process of the present invention. The cartographic data is
16 representative of geographical points along a coastline, wherein
17 each point represented by the data covers a given landmass area.

18 The points are generally indicative of points along the landmass
19 where direction changes occur. The standoff distance is
20 preferably selected by a system operator and represents a
21 distance which is usually sufficient for the avoidance of radar
22 and/or defenses originating from the landmass adjacent which the
23 projectile is to fly, allowing for navigation error and for

1 meeting other rules of engagement. The standoff distance may be
2 further affected by environmental conditions such as inclement
3 weather and the like and as such may be adjusted by the operator.

4 The process of the present invention functions to calculate
5 landmass fly-around boundaries relative to various types of
6 landmasses so that missiles or the like can be maintained in an
7 automated fashion at the desired standoff distance.

8 The fly-around boundaries can be displayed on a video screen or
9 other display device (not shown) associated with computer 10 for
10 use by the operator to define, redefine or adjust the boundaries.

11 Information calculated by the boundary generation process of the
12 present invention may loaded into a projectile navigational
13 control system and used to determine missile presets or launch
14 sequence plans. The navigation control system and the manner in
15 which the missile presets or launch sequence are determined do
16 not form part of the present invention.

17 The first tier of the landmass boundary process of the
18 present invention, which is implemented with the assistance of
19 computer 10, is the sub-process 14 shown in the flow chart
20 illustrated in FIGS. 2A and 2B. This part of the process uses
21 original cartographic or landmass data from device 12 which
22 defines the world vector shoreline (WVS) which is digitized via a
23 digitizer (not shown) into a format including latitudinal and
24 longitudinal coordinates of geographical positions. The
25 digitized cartographic data for particular types of landmasses

1 are categorized in data files in device 12 for retrieval by
2 computer 10 and stored in memory 11 for by sub-process 14 of
3 FIGS. 2A and 2B. By way of sub-process 14, cartographic data is
4 analyzed and processed for removing points therefrom which are
5 not required.

6 Points which are not required are determined by the shape of
7 the coastline, wherein points of data representing small harbors
8 or other inlets along the coastline or the like are removed via
9 sub-process 14 based on their smaller size relative the general
10 configuration of a larger section of landmass to which they
11 belong. This is discussed more specifically below.

12 In step 16, the original cartographic data is accessed for
13 the geographic area of interest. In step 18, the process is
14 initiated. Since the landmass data for the geographic area of
15 interest is processed in small pieces, step 18 is actually a test
16 to determine if the process is complete. This process continues
17 until all landmass harbor and cross-overs are removed. In step
18 20, the points representing the digitized cartographic data are
19 processed. Referring to the schematic representation of a
20 coastline model shown in FIG. 3A, three successive points at a
21 time including a starting end point (1st point), e.g., point A,
22 an intermediate point (2nd point), e.g., point B, and a finishing
23 end point (3rd point), e.g., point C, are analyzed in a
24 sequential, preferably clockwise manner beginning with an entry
25 point for the landmass model, e.g., point A, and ending with an

1 exit point of the model, e.g., point K, in this step. During the
2 sequential processing of the points, the intermediate point
3 becomes the starting end point for the next three points unless
4 it is removed as discussed below and the finishing end point
5 becomes a starting end point in the next set of three points
6 analyzed in the sequence. The three points are used in step 22
7 to determine if the landmass section represented by the three
8 points is a first or second type of configuration, i.e. a concave
9 configuration or a second type convex configuration,
10 respectively.

11 In categorizing the configuration of the landmass
12 represented by three points, and moving in a clockwise direction,
13 for every three points and the segments defined thereby, right
14 hand turns are considered convex and left hand turns are
15 considered concave. Accordingly, and referring to FIG. 3A,
16 following points A, B and C sequentially, the path represented by
17 these points defines a right hand turn while the path represented
18 by points B, C and D defines a left hand turn.

19 Accordingly, and referring by way of example to FIG. 3A,
20 wherein coordinates of actual geographical points along a real
21 coastline are shown, every three points along coastline model 200
22 are in either a convex or concave directional configuration. The
23 concavely configured sections include potentially removable small
24 harbors and the like.

1 Points A, B and C represent a right hand turn and a convexly
2 configured section of landmass; points B, C and D represent a
3 left hand turn and a concavely configured section of landmass;
4 points C, D and E represent a right hand turn and a convexly
5 configured section of landmass; points D, E and F represent a
6 left hand turn and a concavely configured section of landmass;
7 points E, F and G represent a left hand turn and a concavely
8 configured section of landmass; points F, G and H represent a
9 right hand turn and a convexly configured section of landmass;
10 points G, H and I represent a right hand turn and a convexly
11 configured section of landmass; points H, I and J represent a
12 left hand turn and a concavely configured section of landmass;
13 and points I, J and K represent a right hand turn and a convexly
14 configured section of landmass.

15 In step 24, if the three points and the segments thereby
16 represent a convex configuration, for example segments AB and BC
17 defined by points A, B and C, the process is returned back to
18 step 18. Such is the case for segments AB and BC. If, however,
19 the segments defined by the three points represent a concave
20 configuration, for example, as do segments BC and CD defined by
21 points B, C and D, the process moves on to step 26.

22 For every three points where the first and third points are
23 separated by less than the standoff distance, the fly around
24 boundaries which would be generated therefor without removal of
25 the second point from the landmass model, would include very

1 small concave configurations detouring minimally from the
2 remaining portion of the boundary. Accordingly such harbors are
3 removed to avoid such minimal detours and simplify the would be
4 fly around boundary developed from the landmass model, because
5 the increase in distance over the required offset from the
6 landmass model is minimal. It should be noted that if a convex
7 point (i.e., harbors not meeting the criterion is removed when it
8 should not have been removed, the resultant fly-around boundary
9 would be generated more than the stand off distance away from the
10 original land and fly-around boundary will not be accurate and
11 thus the profile of the land mass will not be accurate.

12 The sections of coastline such as that represented by points
13 B; C and D in FIG. 3A are examined in step 26 to determine
14 whether the intermediate or vertex point formed between the two
15 segments, i.e. point C, can be removed for reducing the overall
16 length of the fly-around boundary. If the distance from the
17 starting end point, for example end point B, to the finishing end
18 point, for example end point D, is less than the predetermined
19 standoff distance, the vertex or intermediate point, for example
20 point C, is eliminated in step 28, thus updating the original
21 cartographic data.

22 FIG. 3A is further illustrative of the process represented
23 by steps 24-28 for removing points from coastline model 200. The
24 removal of the intermediate or vertex point does not alter the
25 accuracy of the fly-around boundary since the resulting boundary

1 would encompass the concave section of coastline anyway. After
2 points B, C and D, the process begins again at step 18 moving on
3 to the next set of three points, D, E and F since point C was
4 removed.

5 In FIG. 3A, direct distance D1, D2 and D3 represent the
6 distances which are compared against a predetermined value,
7 preferably the standoff distance. For example, for points B, C
8 and D, the distance D1 is less than the standoff distance stored
9 in memory 11 and accordingly, intermediate point C is removed,
10 thus updating the digitized data representing the coastline,
11 forming new data and beginning the formation of the resultant
12 coastline model 202, shown in FIG. 3B. The new data is stored in
13 memory 11, shown schematically in FIG. 1, and by operation of
14 step 28, the new data is connected by segments, thus
15 reconfiguring the landmass model and creating a resultant
16 landmass model.

17 FIG. 3B represents a resultant coastline or landmass model
18 202. Intermediate points C, E and I were removed from coastline
19 model 200 based on step 26 wherein the distances D1, D2 and D3
20 were less than the standoff distance. The resultant coastline
21 model 202 is represented by points A, B, D, F, G, H, J and K,
22 wherein these points are also connected by segments via the
23 process step 28. Convexly configured sections of the coastline
24 are not removed or altered. After the completion of the analysis
25 of the configurations of all sections of coastline model 200

1 represented sequentially by three points, in step 30, the data
2 representing the resultant coastline or landmass model 202, as
3 shown in FIG. 3B, with small harbors and the like removed, is
4 accessed for the initiation of a cross-over check in step 32 of
5 FIG. 2B.

6 Cross-overs can occur when points are removed from the
7 coastline model 200 and new segments are added to connect the
8 remaining points. Cross-overs are created by segment
9 intersections. Referring to FIGS. 3A and 3B, an example of this
10 situation occurs when point E is removed, which further removes
11 original segments DE and EF from coastline model 200. The
12 landmass configuration defined by segments DF and FG, and FG and
13 GH in FIG. 3B are now convex. However, segment DF and segment GH
14 intersect forming a polygon including portions of segments DF and
15 GH and segment FG.

16 In steps 34, 36 and 38 the cross-over check involves
17 comparing each segment remaining in resultant coastline model 202
18 with all other segments of resultant coastline model 202 for
19 determining any non-endpoint intersections. Two successive data
20 points are used as input in step 36 which are representative of
21 one line segment such as segment DF. Segment DF is preferably
22 compared with all other segments of resultant coastline model 202
23 such as, for example, with segment GH in step 38. In step 40,
24 the process is directed back to step 34 if no intersection is
25 found. If an intersection is found, the process is directed to

1 step 42 wherein a new common point Z, shown in FIG. 3C, is added
2 to resultant coastline model 202 at the non-endpoint intersection
3 point between the intersecting segments DF and GH. In step 42,
4 the intersection point Z becomes a new common end point of the
5 intersecting line segments. In step 44, except for the new
6 common end point Z, as shown in FIG. 3C, all points and segments
7 on the polygon formed originating from point Z and directed away
8 from water side W and toward land side L, are removed.

9 Accordingly, FIG. 3C is representative of a near filtered
10 coastline model defined by points A, B, D, Z, H, J, and K wherein
11 all initial small harbors and cross-overs have been removed.

12 If a new common end point at a line segment intersection was
13 added in process steps 36-44, in step 46, the process is directed
14 to step 48 for accessing the near filtered coastline data as
15 illustrated by the near filtered landmass model in FIG. 3C, with
16 the new common end points and small harbors removed, for
17 reprocessing through the entire process starting with step 18, as
18 described above to recheck for small harbors and cross overs.
19 The repetition of the process starting from step 18 is preferable
20 since removing points and adding new common end points as
21 performed in steps 36-44 alters the coastline configuration and
22 creates new concave/convex coastline configurations.

23 A filtered coastline model 204 is achieved including points
24 and segments when all points are processed without small harbors
25 and cross-over conditions existing, as shown in FIG. 3C. Data

1 representing filtered coastline 204 is stored in the memory 11 of
2 the computer 10, shown in FIG. 1. The filtered data representing
3 filtered coastline model 204 can now be accessed from memory 11
4 and used as a reference for offsetting fly around segments and
5 generating fly-around boundaries.

6 In sub-process 50 implemented by computer 10, represented by
7 the flow charts shown in FIGS. 4A, 4B and 4C, fly-around
8 boundaries are generated. The purpose of creating fly-around
9 boundaries is to generate a boundary around each filtered
10 coastline or landmass at an operator's specified predetermined
11 value or standoff distance at which a projectile such as a
12 missile or the like is to be maintained in flight. The filtered
13 landmass data representative of filtered coastline model 204
14 derived from sub-process 14 discussed above is used for sub-
15 process 50.

16 In step 52 of FIG. 4A, filtered landmass or coastline data
17 having new common end points and small harbors removed is
18 accessed from the memory 11 of the computer 10, shown
19 schematically in FIG. 1. Boundary generation sub-process 50 is
20 applicable to all types of landmasses including multiple point
21 islands, multiple point non-island landmasses, single point
22 islands and line segment islands. For the purpose of this
23 description, a multiple point island landmass is defined as an
24 island which can only be defined by a plurality of non-linear
25 points, given the area of land mass in which one point defines

1 the landmass; a multiple point non-island landmass generally
2 defines a section of non-island coastline; a single point island
3 landmass is defined as a relatively small island which can be
4 defined by one point, given the area of land mass in which one
5 point defines the landmass; and a line segment island landmass
6 typically defines an elongated very narrow island having a width
7 definable by single points, given the area of landmass in which
8 one point defines the landmass. For the latter two types of
9 landmasses, a specialized portion of sub-process 50 is used for
10 generating the fly-around boundaries.

11 In step 54, the process is directed to step 56 if a single
12 point island is being analyzed or the process is directed to step
13 58 if a line segment island or other type of landmass is being
14 analyzed. A line segment island is defined by 3 consecutive
15 points where the first point and the third point are identical.

16 In step 56, if only a single point representative of a
17 single point island is being analyzed as shown in FIG. 5A, a
18 polygon preferably having equal length connected sides,
19 preferably in the form of an octagon 210, is preferably generated
20 around the single digitized point A, as illustrated in FIG. 5A.
21 The octagon is generated with the single landmass point in the
22 center. The connected sides of octagon 210 are representative of
23 original fly-around segments, and the center point C of each side
24 or fly around segment is separated from the single point at a
25 distance equal to the standoff distance S.

1 To generate the octagon, the eight endpoints P are
2 established around center point C using the standoff distance S
3 and the angle 2θ representing the angle between the endpoints of
4 each side of the octagon, where the center point C of each side
5 is offset the standoff distance S from center point A.
6 Accordingly, angle 2θ is known at 40° , wherein θ equals 20° .
7 From these known parameters, the radius R can be determined by
8 the formula $R = D/\text{COS}\theta$. The points can be generated for octagon
9 210 with point A as the origin and starting at 0° . A first point
10 P1 is plotted first at 20° at the end of the known radius R.
11 Seven more points P are plotted 40° apart, each at the end of the
12 known radius R.

13 Alternatively, if the landmass is not a single data point,
14 the process is directed to step 58. If the filtered landmass
15 data or model constitutes a line segment island, as defined
16 above, such as that shown in FIG. 5B, a fly-around boundary is
17 preferably created in step 60 by generating two octagons 212 and
18 214, as discussed above for FIG. 5A. Octagons 212 and 214 have
19 preferably equal length connected sides representative of
20 original fly-around segments and are generated around each
21 endpoint A and B of the line segment island, wherein as before
22 the center points C of each side of each octagon is located the
23 standoff distance S from the endpoint. The octagons are
24 connected with additional fly-around segments CD and EF placed

1 substantially parallel to the periphery of the line segment
2 island. Points and segments or portions' thereof comprising the
3 sides of the octagons, as shown by the dotted lines in FIG. 5B,
4 and located between the additional fly-around segments, are
5 eliminated. If the filtered landmass model does not constitute a
6 line segment island, the process is forwarded to step 62, as
7 shown in FIG. 4B.

8 In step 62, if additional data exists for processing, i.e.,
9 boundary generation is not complete due to (1) the data not being
10 representative of a line segment island or a single point island,
11 or (2) at least one more set of three points exists in the
12 sequence of points for processing, the process moves to step 64.

13 Otherwise the process moves to step 82, discussed below.

14 The process beginning with step 64 analyzes three successive
15 points at a time of the filtered landmass model 204, as discussed
16 above in detail for sub-process 14, wherein the three successive
17 points represent either a convex or concave landmass
18 configuration. In step 66, therefore, based on the three points
19 obtained in step 64, the process determines whether or not the
20 section of landmass represented by the three points from the
21 filtered landmass model forms a concavely or convexly configured
22 section of the coastline. The process of step 66 is the same as
23 that described above for process 14 and shown in FIG. 2A, step
24 24.

1 In step 68, to define the fly-around boundary, infinitely
2 long line segments are offset in a parallel orientation from the
3 segments of filtered coastline model 204. Each infinitely long
4 line segment intersects with the infinitely long line segment
5 offset from an adjacent filtered coastline model segment. For
6 convex configurations, the portions of the infinitely long
7 segments extending past the intersection point with adjacent
8 infinitely long segments are removed in steps 72-76. For concave
9 configurations, the portions of the infinitely long segments
10 extending past the intersection point with adjacent infinitely
11 long segments are removed as discussed below in step 78.

12 Steps 68-76 are illustrated in FIGS. 6 and 7, wherein
13 original fly-around segments EF and FG are offset from segments
14 AB and BC, respectively, of filtered landmass or coastline model
15 204. In step 68, fly-around segments EF and FG are offset in a
16 sequential manner from segments AB and BC toward water side W a
17 distance equal to the standoff distance S. As an example of this
18 sequential manner, each set of two segments includes a starting
19 segment, e.g., AB, and a finishing segment, e.g., BC, wherein the
20 finishing segment BC becomes the starting segment in the next set
21 of segments subject to offset, e.g., segments BC and CD, having
22 original fly around segments FG and GH offset therefrom, as shown
23 in FIG. 7. The water side W of the segments of the filtered
24 coastline model is on the left side of the segments when viewing
25 points sequentially in a clockwise direction.

1 From step 68 the process is directed to step 70 whereat if a
2 convex configuration was found in step 66, the process is further
3 directed to step 72. If not, the process is directed to step 78,
4 discussed below.

5 For landmasses with convex configurations or coastlines as
6 shown in FIGS. 6 and 8, an error results in the offsetting of the
7 original fly-around segments from the segments at the apex area
8 of the convex configurations of the fly around boundaries, and
9 accordingly, the fly-around boundary must be adjusted. The
10 resulting error can be described with reference to FIG. 8.

11 FIG. 8 shows a filtered section of landmass with convex
12 configuration model defined by points A, B and C and the
13 adjoining segments AB and BC, wherein point B is the vertex. The
14 fly-around boundary is represented by points E, F and G and the
15 adjoining original fly-around segments EF and FG, wherein point F
16 is the initial vertex of the fly-around boundary, wherein
17 portions of segments EF and FG extending toward the water beyond
18 vertex point F are eliminated. Point F of the fly-around
19 boundary is offset from vertex B a distance d greater than the
20 standoff distance S , an error inherently associated with the
21 offsetting process. Distance d represents an offset error,
22 which, if not corrected, will result in an unnecessarily
23 lengthened fly-around boundary. For convexly configured
24 landmasses, the error increases as the angle α between the
25 segments becomes more acute. The following adjustments must be

1 performed on convexly configured landmasses or sections of
2 coastline beginning at step 72.

3 In step 72, a line segment is defined which bisects the
4 angle α formed by the convexly arranged landmass segments AB and
5 BC, extending a distance equal to the standoff distance from
6 vertex point B, as defined by segment BH in FIG. 8. In step 74,
7 a line segment perpendicular to segment BH and offset a distance
8 equal to the standoff distance from the vertex B is provided as
9 defined by segment IJ, which also intersects with original fly-
10 around segments EF and FG, respectively. Referring to step 76
11 and FIG. 8, the initial fly-around boundary vertex point E and
12 portions of original fly-around segments extending beyond segment
13 IJ, i.e. segments IF and FJ, are eliminated and replaced by new
14 fly-around segment IJ, creating a new fly-around boundary.
15 Accordingly, points E, I, J, and G in FIG. 8 denote the new fly-
16 around points and new fly-around boundary after the error is
17 corrected. For concavely configured landmasses or sections of
18 coastline, the fly-around vertex point is used without
19 modification as there is no error introduced via sub-process 50.

20 Because of the addition of extra data points, for example,
21 points H and I, for convexly configured landmasses or sections of
22 coastline necessary for the elimination of standoff distance
23 error, the number of points in the landmass in which the fly-
24 around vertex error has been corrected, is increased. Because of
25 the additional points, new errors can arise, wherein the

1 additional points can violate the standoff distance offset
2 requirements of segments of adjacent landmass configurations.
3 The new error is preferably corrected as set forth beginning in
4 step 78. Such errors occur when consecutive sets of segments
5 representing filtered landmass models or coastline sections
6 switch from concave to convex configurations or vice-versa.

7 In these situations, the originally offset computed fly-
8 around boundary may need adjustment to adhere to the standoff
9 distance S requirements for adjacent landmass configurations.
10 With reference to FIG. 9 and step 78 of FIG. 4C, if the adjacent
11 landmasses do switch from concave to convex or vice-versa, the
12 process is forwarded to step 80 where the adjustments to the fly-
13 around boundaries are performed in accordance with the diagram of
14 FIG. 9 and the infinitely long fly around segments are trimmed at
15 their intersection points to segments AB and BC. For the
16 convexly configured landmass or coastline section represented by
17 segments AB and BC, and defined by points A, B and C and for
18 concavely configured landmass or coastline section represented by
19 segments BC and CD, and defined by points B, C and D, steps 64-
20 76, as described above, are performed. That is, as shown by FIG.
21 8, original fly-around segments EI, IJ and JG are calculated for
22 convex landmass ABC via steps 70-76, forming landmass model
23 boundary EIJG. Original fly-around segment KL is then offset the
24 standoff distance from segment CD of concave filtered landmass
25 model section BCD.

1 As shown in FIG. 9, point J and an additional portion of
2 segment IJ of boundary EIJG violates the required predetermined
3 value or standoff distance offset from segment CD. Original
4 fly-around segment KL intersects with original fly-around segment
5 IJ of the fly-around boundary for filtered landmass model section
6 ABC at point J'. Point J' does not violate the required standoff
7 distance for segments BC or CD of the adjacent convex and concave
8 filtered landmass model sections. The portions of fly-around
9 boundary EIJG which violate the standoff distance S for segment
10 CD are eliminated. Accordingly, segments J'J and JG are
11 eliminated. In addition, segment KJ' is eliminated for it
12 contains points which violate the standoff distance for segment
13 BC. Thus, the new fly-around boundary for the landmass
14 represented by points ABCD becomes boundary EIJ'L. In some
15 cases, the geometry of adjacent convex/concave landmass segments
16 will result in the fly-around boundary for the second segment of
17 the concave segment, i.e., original fly-around segment KL for
18 segment CD, encompassing both convex fly-around points, for
19 example and not shown, points I and J. When this situation
20 occurs fly-around point J' replaces both convex landmass fly-
21 around points since both points would violate the fly-around
22 boundary for segment CD. For segments of filtered landmass model
23 or coastline sections changing from concave to convex
24 configurations, the same fly-around adjustments as discussed
25 above may be required.

1 In step 81, the process is sent back to step 62 continuing
2 through this loop until all data has been processed in accordance
3 with steps 64-80, assuming the special conditions of steps 54-60
4 do not exist. If the data has been entirely processed, i.e., no
5 new sets of three points are available, the process moves to step
6 82 whereat if the landmass is a multi-point island landmass,
7 processing is ended. A multi-point island landmass is determined
8 by the process when both the entrance and exit points of the
9 landmass are the same. In step 84, however, if the landmass is a
10 portion of a non-island landmass as shown in FIG. 10, including
11 an entry point A and an exit point E, exit and entry original
12 fly-around segments AB and DE coinciding with entry and exit fly-
13 around points, A and E, respectively, are preferably extended.
14 The exit and entry original fly-around segments AB and DE are
15 extended outwardly a distance equal to the standoff distance, as
16 shown in FIG. 10. The extension is done to estimate the fly-
17 around boundary for the portion of the non-island landmass not
18 being currently analyzed.

19 For sub-process 50 described above, the execution time can
20 be reduced at the expense of increased memory allocation and
21 decreased tactical capability. That is, fly-around boundaries
22 could be calculated off-line, i.e., not real time, for a
23 predefined set of distances corresponding to specific operating
24 areas. These predefined distances which set the boundaries can
25 be saved in memory 11 and displayed at a terminal (not shown) of

1 computer 10, at the operator's request. Such a process would
2 restrict the operator from launching missiles from other than the
3 predefined operating areas and from using standoff distances
4 other than the predefined distances.

5 Referring now to FIG. 11, sub-process 86 is disclosed for
6 combining multiple fly-around landmass boundaries which overlap
7 each other, into a single fly-around landmass boundary. Once
8 such multiple overlapping landmass fly-around boundaries are
9 combined, they are utilized by the navigational system shown in
10 FIG. 1 for landmass boundary avoidance. Combining overlapping
11 fly-around boundaries reduces the number of boundaries and
12 boundary data points, which improves operability of the
13 navigational system by decreasing the complexity of the flight
14 path.

15 Overlapping boundaries occur when any points on two
16 different original landmasses boundaries are less than twice the
17 standoff distance from each other. Sub-process 86 is, therefore,
18 used to combine all fly-around boundaries of all landmasses
19 which intersect twice.

20 When intersection is found between the segments comprising
21 the base landmass and remaining landmasses boundaries, a test
22 using the dot product of the intersecting original fly-around
23 segments is preferably used to determine if following the base
24 landmass boundary through the intersection enters or exits the
25 remaining landmass. The intersections are labelled as entry

1 points e1 or exit points e2. The sign of the dot product of the
2 intersecting segments is used to determine whether a right or
3 left hand turn is made, wherein a right turn is an entry and a
4 left turn an exit.

5 It should be noted that the dot product of two vectors is a
6 standard mathematical formula which is defined to be the product
7 of two vector lengths by the cosine of the angle between them.
8 For our use, the dot product provides a simple way to find the
9 angle between the intersecting landmass vectors. To determine if
10 the remaining landmass vector is heading into or out of the base
11 landmass, the intersecting angle between the vectors is measured.

12 By normalizing the base segment vector and rotating it
13 counterclockwise by 90 degrees, the dot product with the next
14 landmass segment will result in either a positive (left turn/exit
15 the base segment) or a negative (right turn/enter the base
16 landmass) value. Determining the order of the points on the new
17 combined landmass boundary and points to be eliminated from the
18 original land mass boundaries is handled differently for each of
19 the following cases.

20 When the boundaries being combined for both the base
21 landmass B and remaining landmass R, are for island landmasses,
22 i.e. closed polygons, as shown in FIG. 12A and they intersect
23 exactly twice, the resultant combined landmass boundary is also
24 in the shape of an island. The combined fly-around boundary is
25 preferably formed by starting from the exit point e2, following

1 the base landmass boundary clockwise to the entry point e1, then
2 following the remaining landmass boundary back to the exit point
3 e2.

4 If the base landmass B is an island type landmass and the
5 remaining type landmass R is a coastal segment, i.e., a non-
6 closed polyline, as shown in FIG. 12B, and the boundaries thereof
7 intersect exactly twice, the resultant combined fly-around
8 boundary is in the form of a coastal segment. The combined
9 landmass boundary is preferably formed by following the remaining
10 landmass boundary from the beginning, continuing along the
11 remaining landmass boundary until the exit point e2, following
12 the boundary of the base landmass to the entry point e1 and
13 finally following the boundary of the remaining landmass to its
14 end.

15 If the base landmass B is for a coastal segment and the
16 remaining landmass R is an island, as shown in FIG. 12C, and the
17 fly-around boundaries thereof intersect exactly twice, the
18 resultant combined landmass fly-around boundary is in the shape
19 of a coastal segment. The combined landmass is preferably formed
20 by starting from the beginning of the base landmass boundary,
21 following the base landmass boundary to the entry point e1,
22 following the remaining landmass to the exit point e2 and finally
23 following the base landmass to its end. The methodology
24 illustrated in FIGS. 12A, 12B and 12C is facilitated by sub-
25 process 86, as discussed below.

1 Step 88 of sub-process 86 acquires data from sub-process 50
2 representing generated fly-around boundaries for a plurality of
3 landmasses such as, for example, island type landmasses 220, 222,
4 224, 226, 228, 230, 232 and 234 shown in FIG. 13A, having fly-
5 around boundaries 320, 322, 324, 326, 328, 330, 332, and 334 as
6 shown in FIG. 13A. Each boundary comprises a plurality of
7 original fly-around segments. Referring to FIGS. 11 and 13, in
8 step 90 a base landmass is designated, for example, landmass 220
9 and data representing the fly-around boundary of landmass 220 in
10 the form of original fly-around segments and points and data
11 representing the fly-around boundaries for all remaining
12 landmasses are also inputted into the process.

13 In step 92, the segments forming the fly-around boundary of
14 the base landmass, for example, boundary 320 is compared with the
15 segments forming the fly-around boundaries of all remaining
16 landmasses, for example boundaries 322-334 for determining if the
17 original fly-around segments comprising the boundary of base
18 landmass 220 intersect any fly-around segments comprising the
19 fly-around boundaries of the remaining landmasses at two points.

20 In step 94, the process is directed to step 96 if an
21 intersection between the original fly-around segments of the base
22 landmass and the original fly-around segments of the remaining
23 landmasses are found at two points.

24 In step 96, a new base landmass and a new fly-around
25 boundary is created from the segments forming the previous

1 intersecting boundaries of the base landmass and remaining
2 landmasses. That is, whenever the base landmass boundary
3 intersects one of the remaining landmass boundaries, the
4 intersecting remaining landmass boundary is incorporated into the
5 base landmass boundary forming a new base landmass boundary 320'
6 as set forth in step 98 of FIG. 11 and as shown in FIGS. 12A,
7 12B, 12C and 13B. Overlapping original fly-around segments and
8 points trapped between the points of intersection, no longer
9 boundary points, are thus eliminated creating the resultant fly-
10 around boundary 320' for landmasses 220 and 222. The resultant
11 fly-around boundary becomes the new base landmass boundary in
12 FIG. 13B in step 98 of FIG. 11. The new base landmass boundary
13 is then sent through the process steps 88 to 100 to again
14 determine if any remaining landmass boundaries intersect with the
15 new base landmass boundary.

16 The process is preferably carried out comparing the
17 segments of the landmass boundary or new base landmass boundary
18 with the segments of the remaining landmass boundaries until
19 either more than two line segments intersect or all the
20 comparisons are complete, resulting in no intersection.
21 Accordingly and referring to FIG. 13B, for example, new base
22 landmass fly-around boundary 320' is sent through sub-process 86
23 for comparison with the fly-around boundaries of the remaining
24 landmasses. In step 94, the segments of boundary 320' is
25 determined not to intersect with segments of remaining boundaries

1 324-334 of landmasses 224-234. The process then proceeds to step
2 100 where the base landmass and its boundary is replaced by the
3 next available landmass and boundary, i.e., landmass 224 and its
4 boundary 324. Referring still to FIG. 13B, boundary 324 is now
5 compared for intersections with the remaining landmass boundaries
6 326-334. Boundary 324 remains unchanged as boundary 324 does not
7 intersect with any other landmass boundaries.

8 In step 100, landmass 226 and boundary 326 are designated
9 the new base landmass and boundary and the process continues
10 checking for intersection between the segments forming fly-around
11 boundary 326 and the segments forming fly-around boundaries 328-
12 334. Boundary 326 is combined with boundary 328, as shown in
13 FIG. 13E, and as described above with regard to boundaries 320
14 and 322, and designated the new base landmass boundary 326' in
15 step 98, wherein the points, original fly-around segments and
16 portions thereof between the points of intersection are
17 eliminated. In step 94, the segments forming base landmass
18 boundary 326' intersect with the segments forming remaining
19 landmass boundary 330, and thus a new base landmass boundary
20 326'' is created in step 96 and designated again in step 98, as
21 shown in FIG. 13D. The data is sent back through the loop and in
22 step 94, the segments of base landmass boundary 326'' are
23 determined to intersect with the segments of remaining landmass
24 boundary 332, and thus a new base landmass boundary 326''' is
25 created in step 96, as shown in FIG. 13E and designated in step

1 98. Finally, the segments of base landmass boundary 326''' are
2 checked for intersection with the segments of landmass boundary
3 334 and no intersection is found in step 94, leaving no more data
4 available for step 88 and ending the process loop. The resulting
5 fly-around boundaries are shown in FIG. 13E.

6 Upon completion of processing landmass boundary 326''', the
7 process for combining fly-around boundaries for the landmasses
8 indicated in FIGS. 13A, 13B, 13C, 13D and 13E is complete because
9 (1) landmasses which have been combined into other landmass
10 boundaries are never designated as a base landmass boundary and
11 (2) the last landmass boundary in the sequence of landmass
12 boundaries has already been checked for intersection with all
13 landmasses boundaries and is not designated as the base landmass
14 boundary.

15 The resultant landmass fly-around boundaries generated can
16 be used in combination with non-intersecting fly around
17 boundaries to achieve a desired flight path. For example,
18 portions of resultant fly around boundaries 320' and 336''' can
19 be used in combination with other non-intersecting fly-around
20 boundaries, coastal or otherwise, for forming a complete flight
21 path for a missile or other projectile.

22 While computer 10 and its sub-processes 14, 50 and 86 are
23 generally applicable to landmasses having oceanic coastlines,
24 using cartographic data representative thereof, the process and
25 sub-processes may also be applicable to inland coastlines

1 adjacent rivers and lakes. In addition, the process may be
2 applicable to inland landmass avoidance if the data is properly
3 presented for use by the process.

4 The process is preferably implemented as a software program
5 written in any applicable programming language. The program can
6 be used by the computer for analyzing the data in accordance with
7 the processes discussed above. Accordingly, each of the numbered
8 boxes comprising the process steps of FIGS. 2A and 2B, 4A, 4B,
9 and 4C and 11 can be performed by lines of code comprising the
10 program.

11 The primary advantage of this invention is that an automated
12 process is provided for determining fly-around boundaries
13 relative to landmasses for use by missiles or the like. Another
14 advantage of this invention is that an automated process is
15 provided for creating fly-around boundaries which functions to
16 optimize a flight path by eliminating small harbors and inlets of
17 coastlines from a model of a landmass coastline. Still another
18 advantage of this invention is that a process is provided for
19 automatically creating fly-around boundaries based on a desired
20 standoff distance from a landmass coastline. And still another
21 advantage of this invention is that a process is provided for
22 creating fly-around boundaries for landmasses, which combines the
23 fly-around boundaries for adjacent intersecting landmass
24 boundaries into a resultant fly-around boundary, depending upon
25 the proximity of the landmasses.

1 It is to be understood that the invention is not limited to
2 the illustrations described and shown herein, which are deemed to
3 be merely illustrative of the best modes of carrying out the
4 invention, and which are susceptible of modification of form,
5 size, arrangement of parts and details of operation. The
6 invention rather is intended to encompass all such modifications
7 which are within its spirit and scope.

2
3 LANDMASS FLY-AROUND BOUNDARY GENERATION

4
5 ABSTRACT OF THE DISCLOSURE

6 A process for generating fly-around boundaries for use by
7 projectiles such as missiles or the like which facilitates the
8 steps of providing original cartographic data representative of
9 at least one geographical position on a landmass and providing a
10 predetermined value for the spacing of a fly-around boundary from
11 the geographical position; digitizing the original data and
12 creating a landmass model in a format which includes latitudinal
13 and longitudinal coordinates of the geographical position;
14 inputting the digitized data and the predetermined value into a
15 data analyzer; generating new data representative of the fly-
16 around boundary using the data analyzer based on the
17 predetermined value and the land mass model, preferably by
18 offsetting fly-around segments from the landmass model a distance
19 equal to the predetermined value; and providing a navigational
20 control for receiving the new data and using the new data for
21 controlling the flight path of the missile along the boundary.

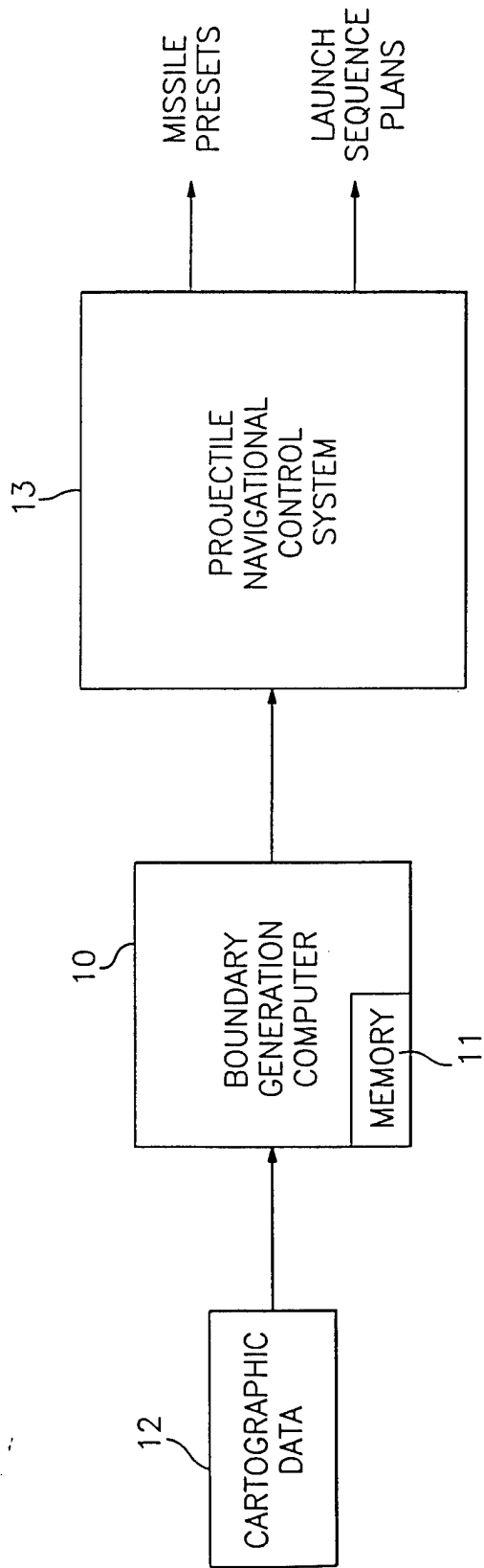


FIG. 1

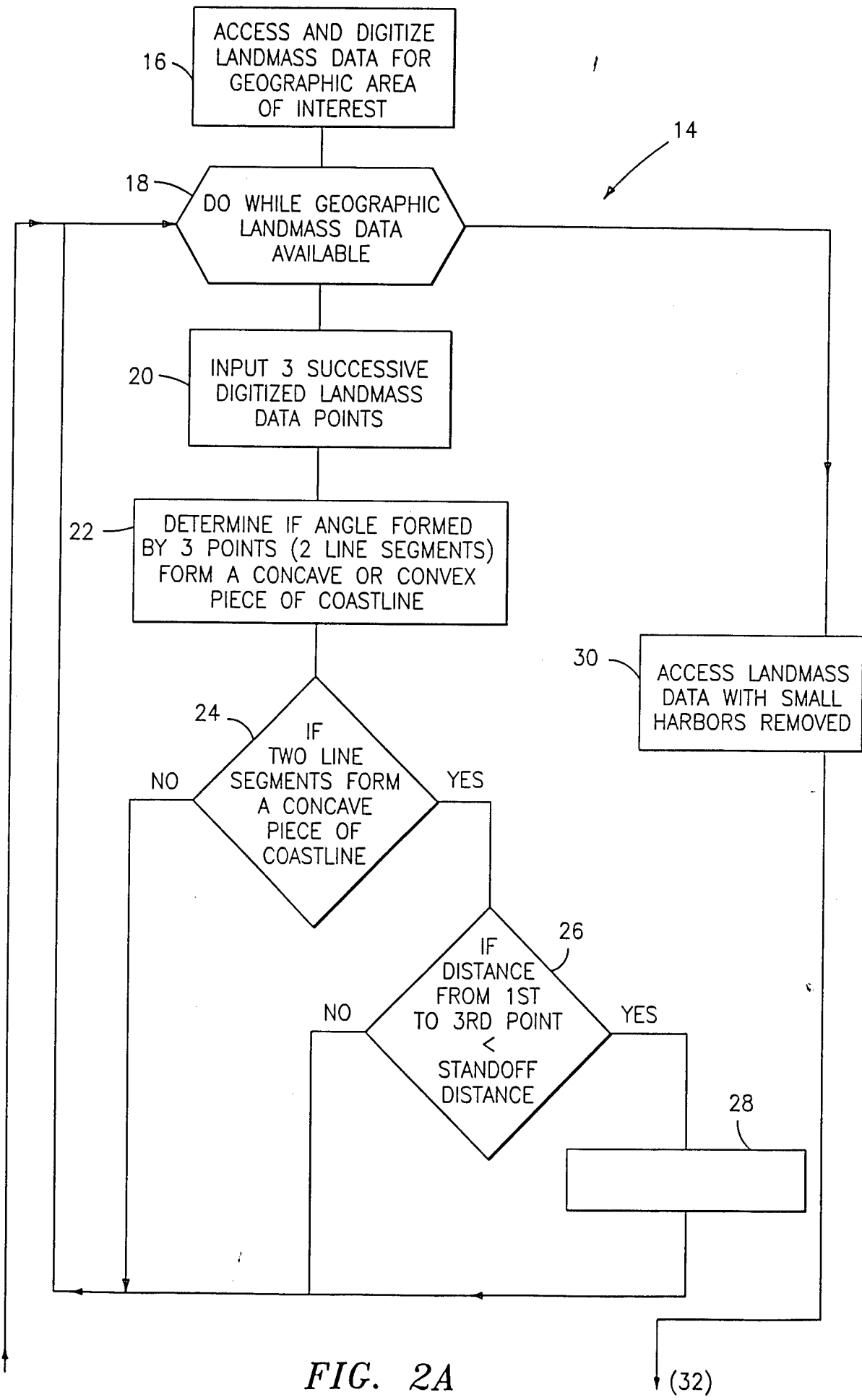


FIG. 2A

(32)

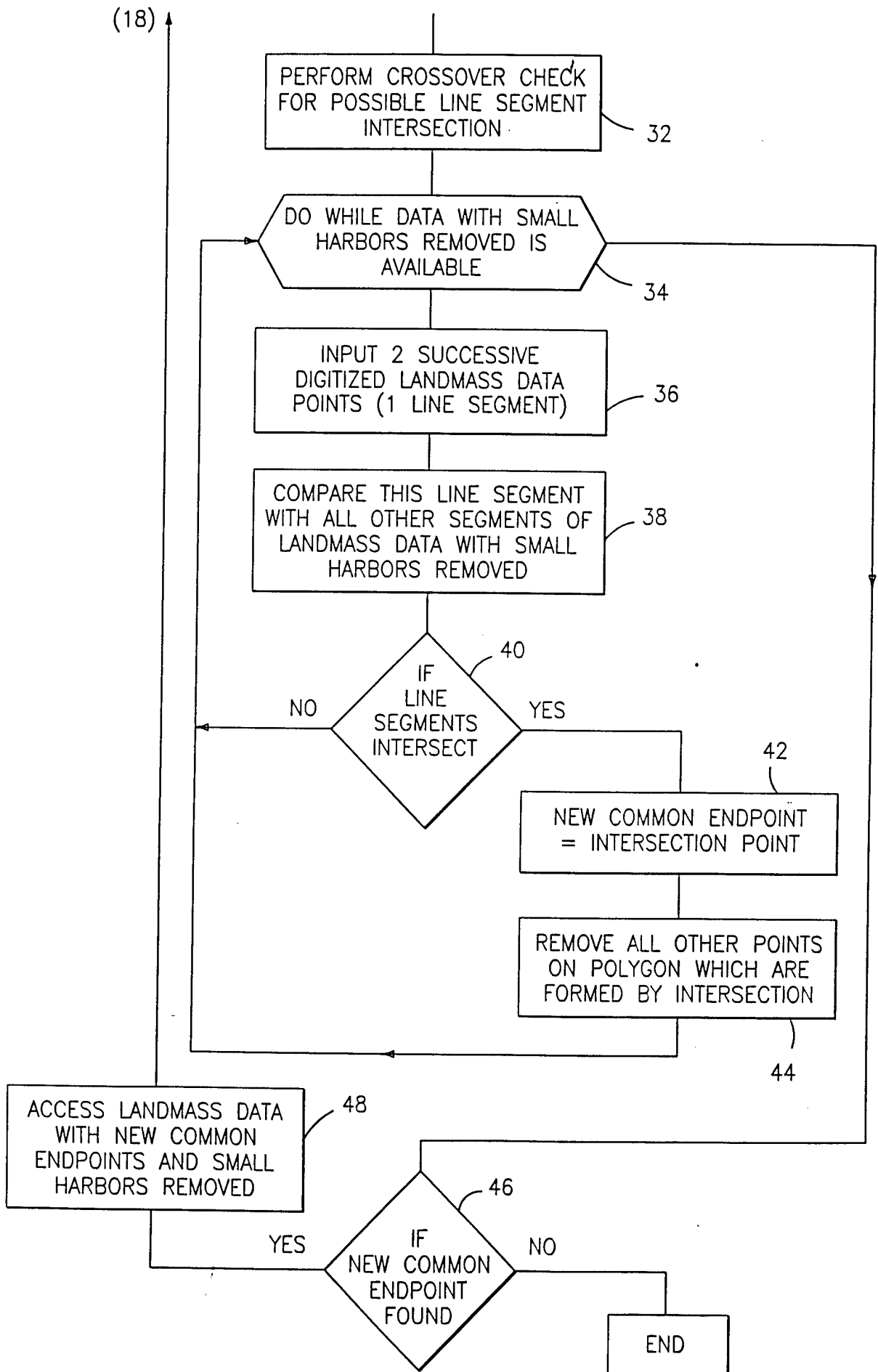


FIG. 2B

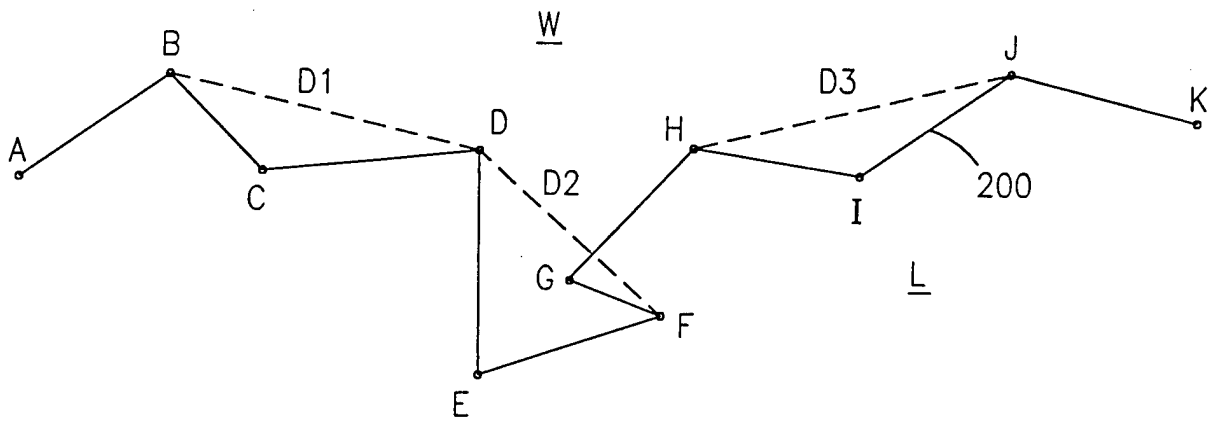


FIG. 3A

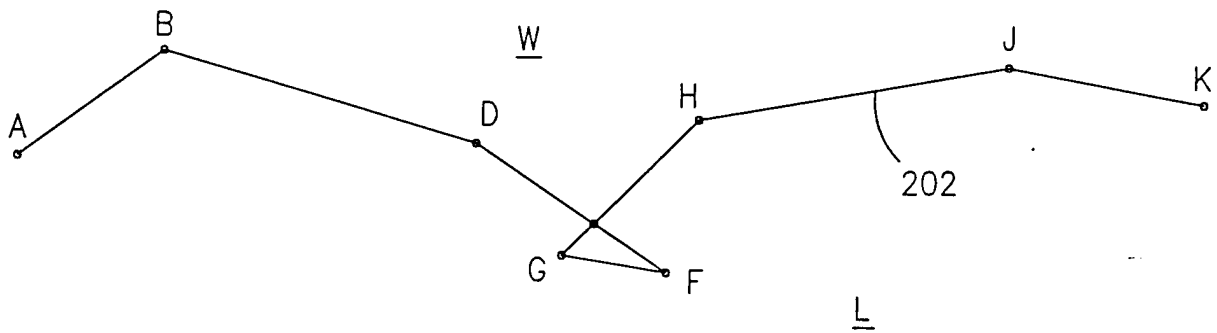


FIG. 3B

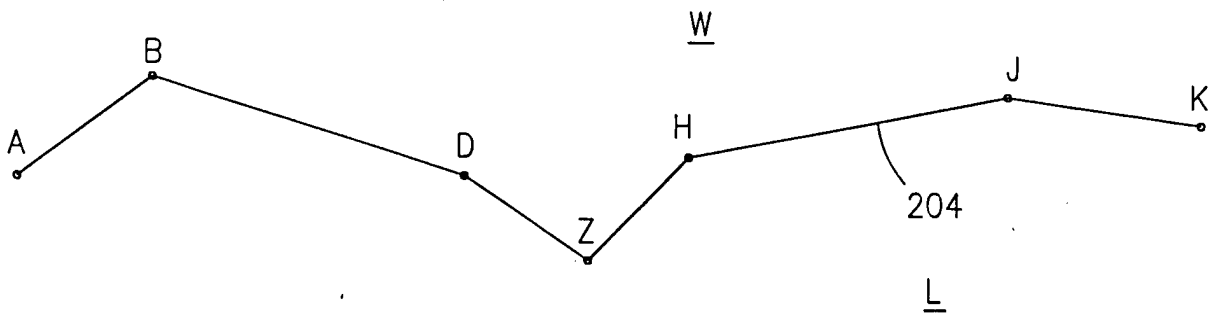


FIG. 3C

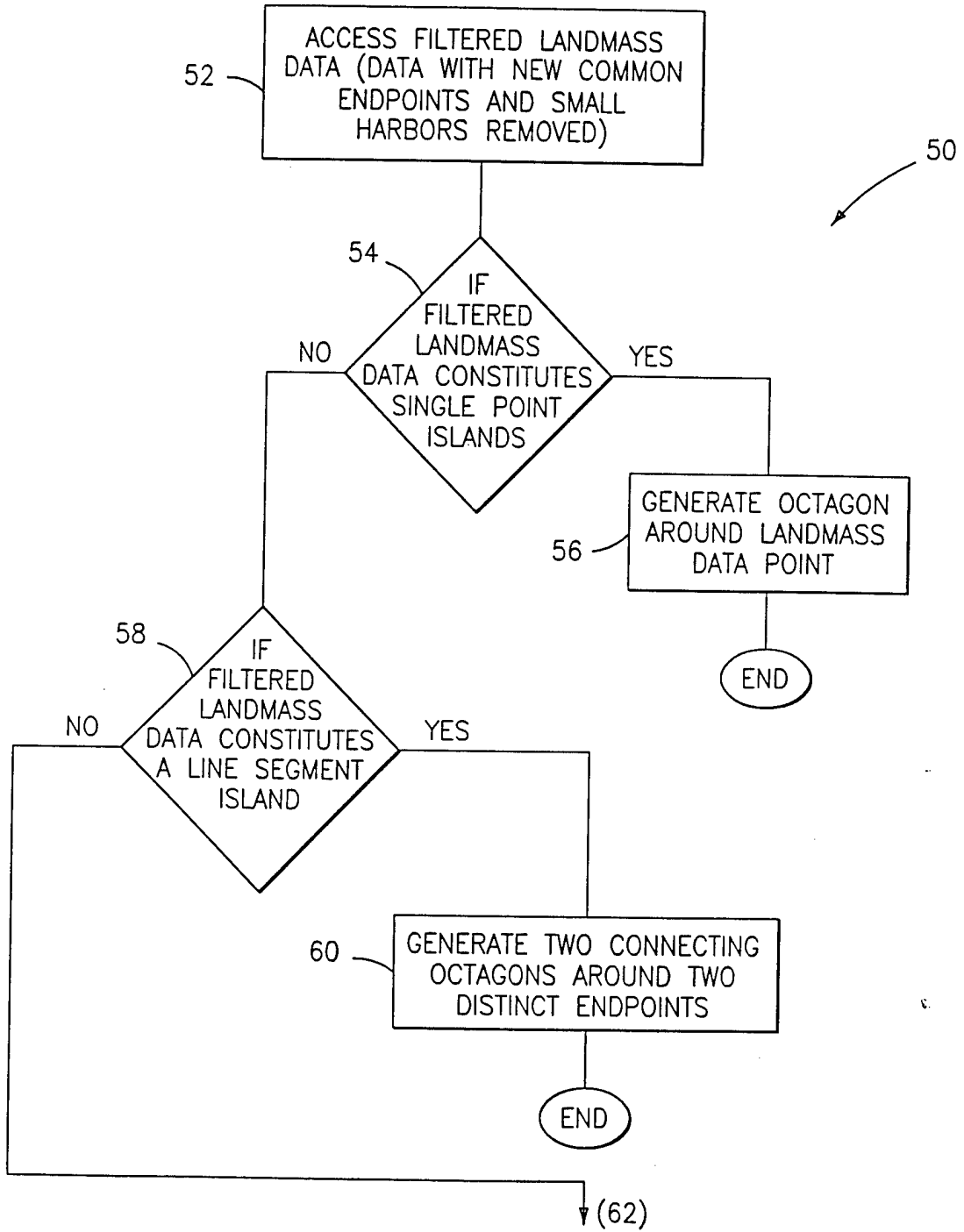


FIG. 4A

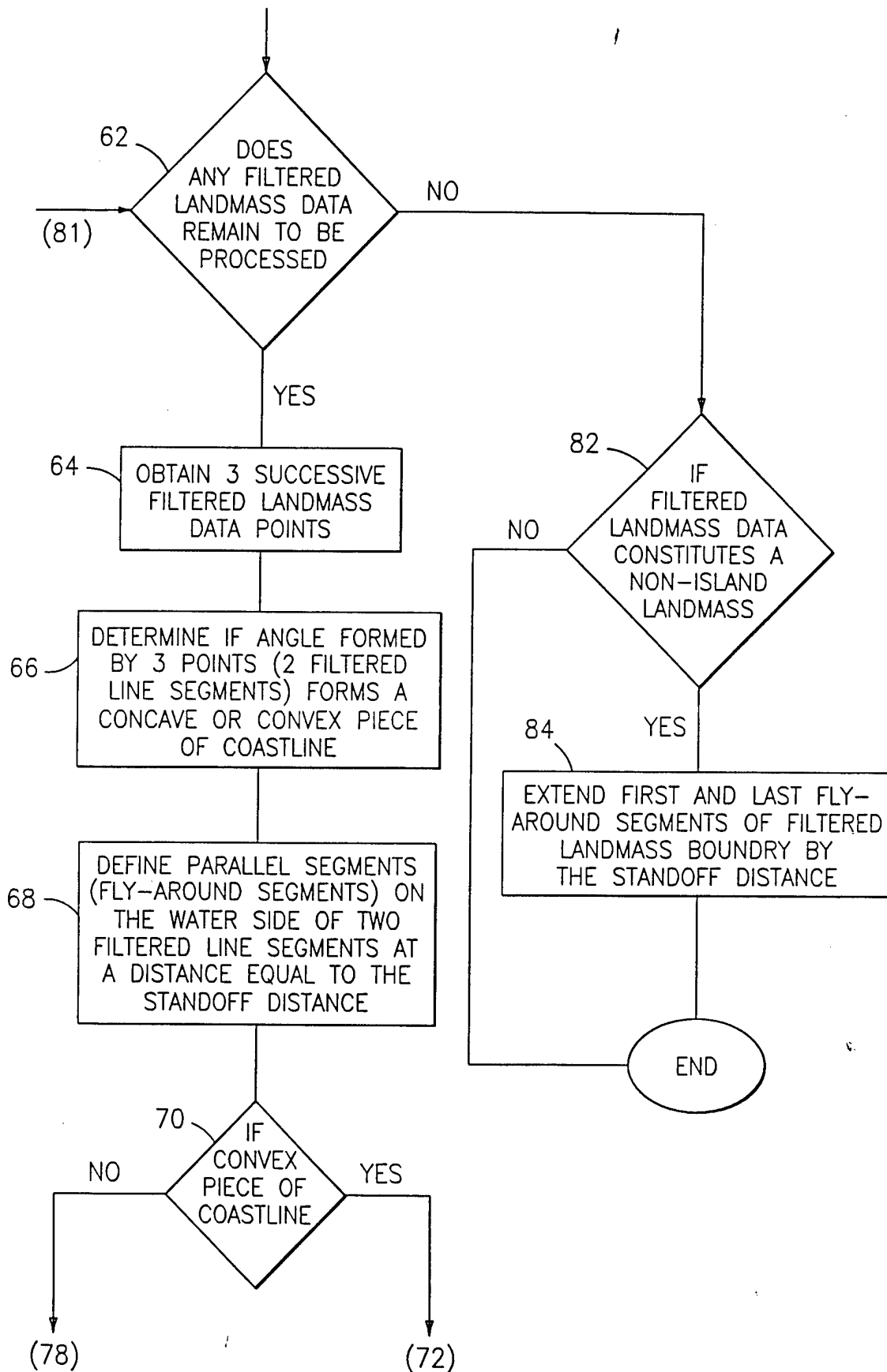


FIG. 4B

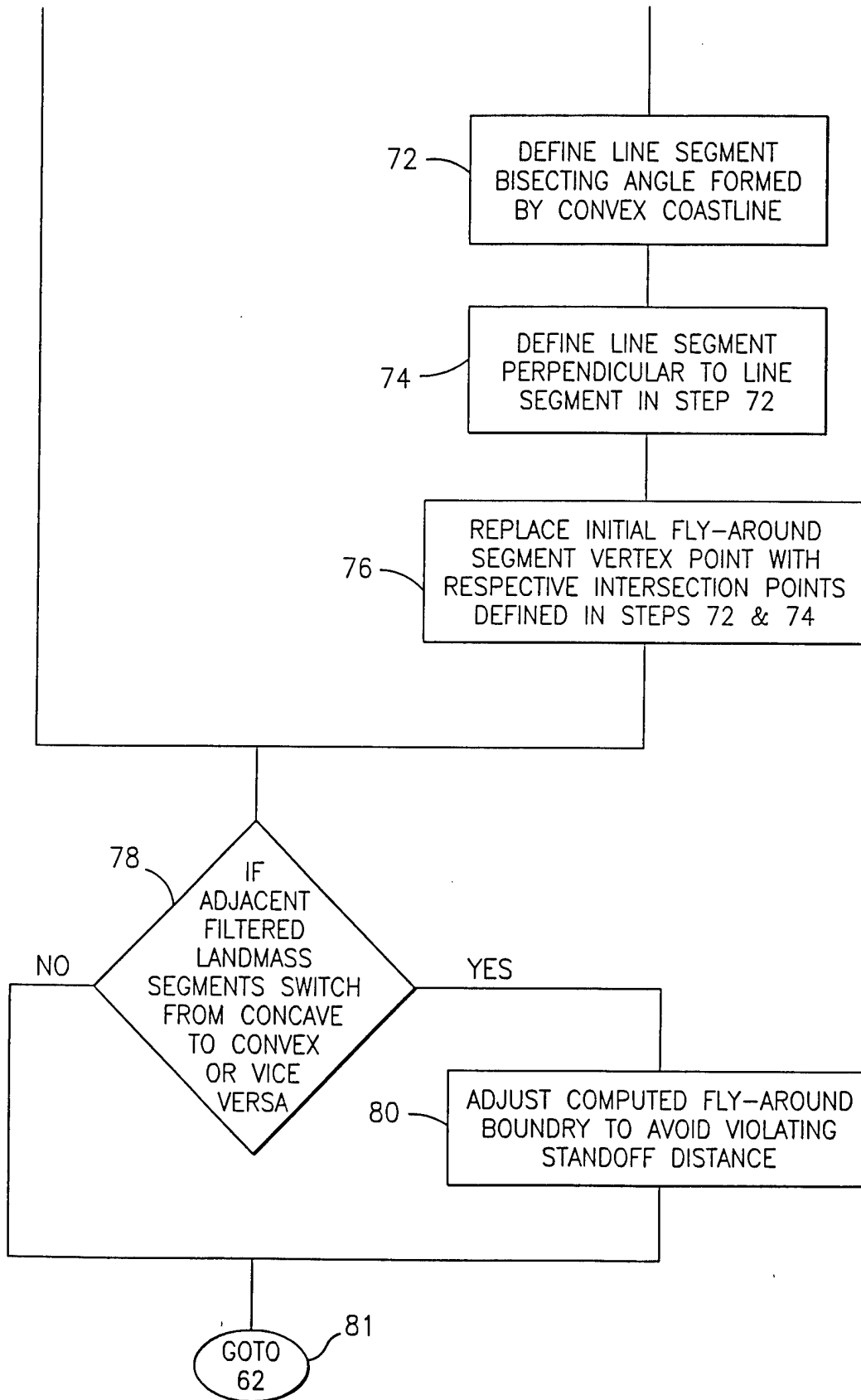


FIG. 4C

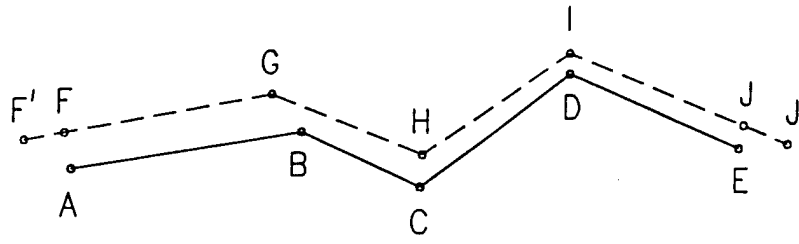


FIG. 10

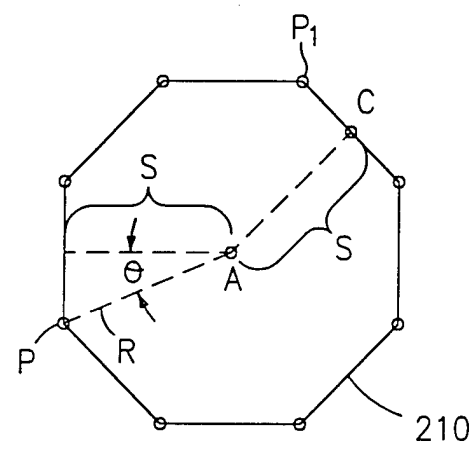


FIG. 5A

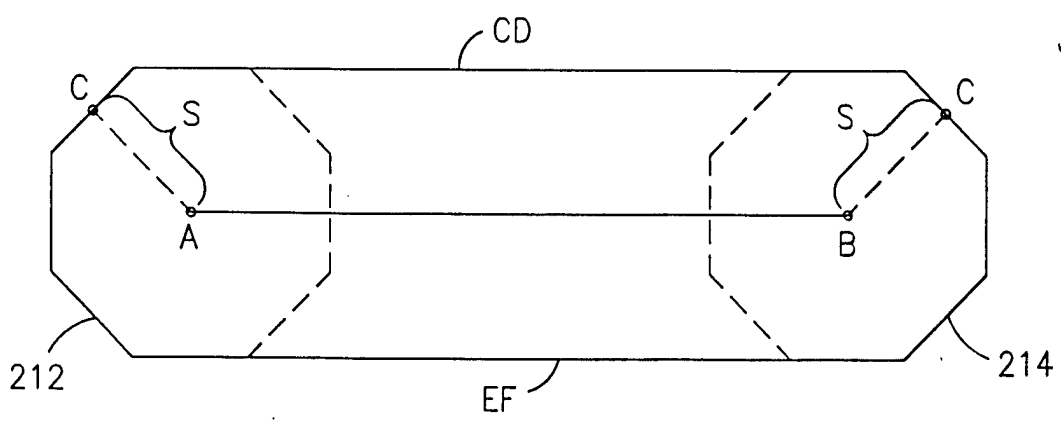


FIG. 5B

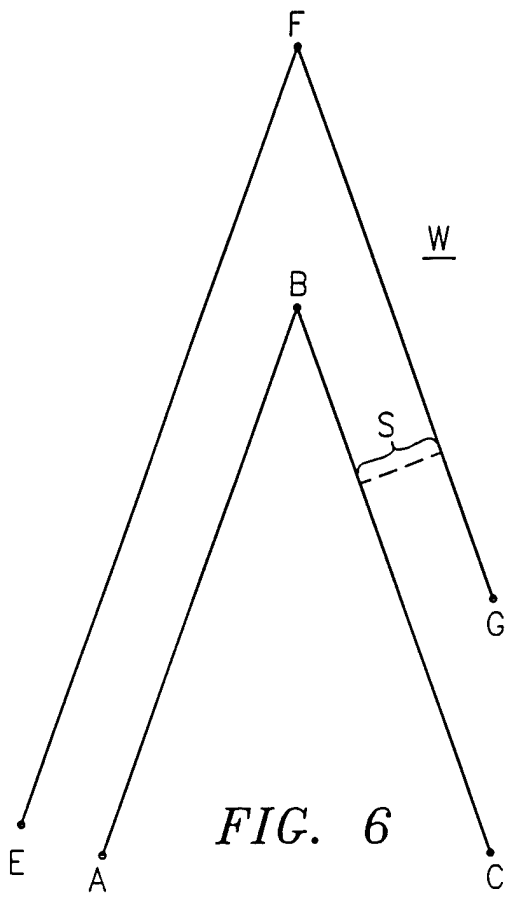


FIG. 6

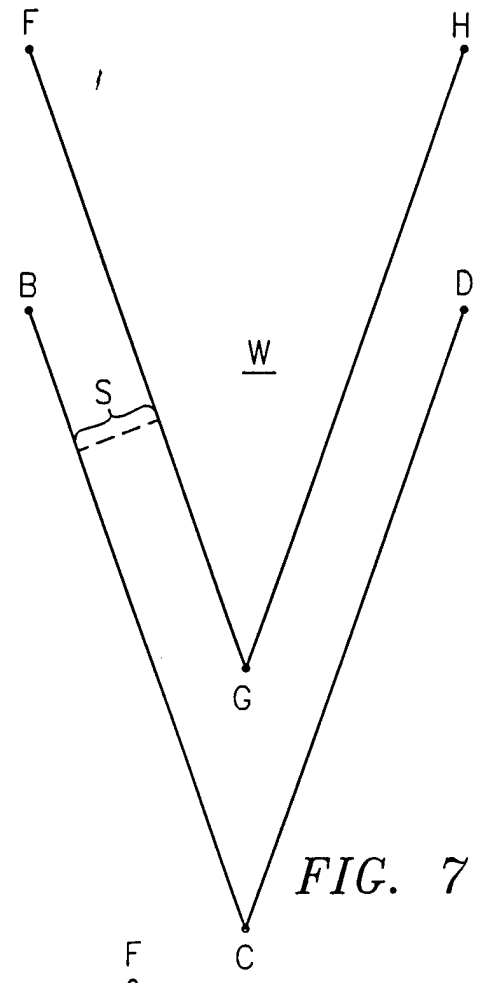


FIG. 7

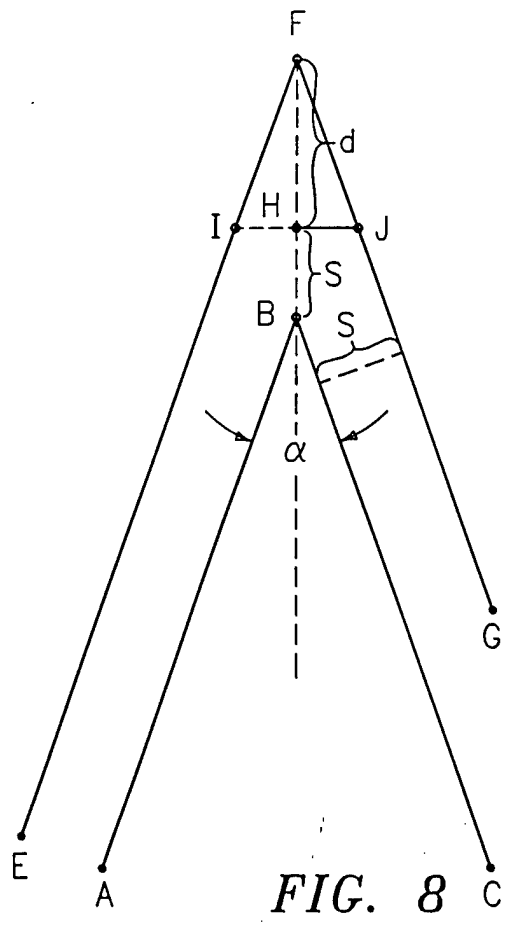


FIG. 8

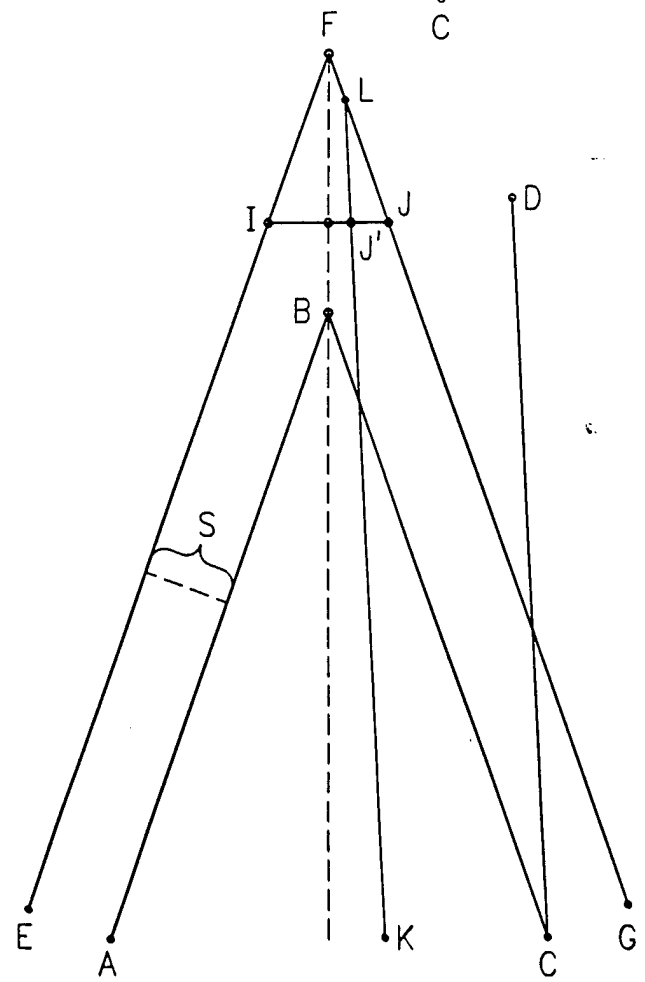


FIG. 9

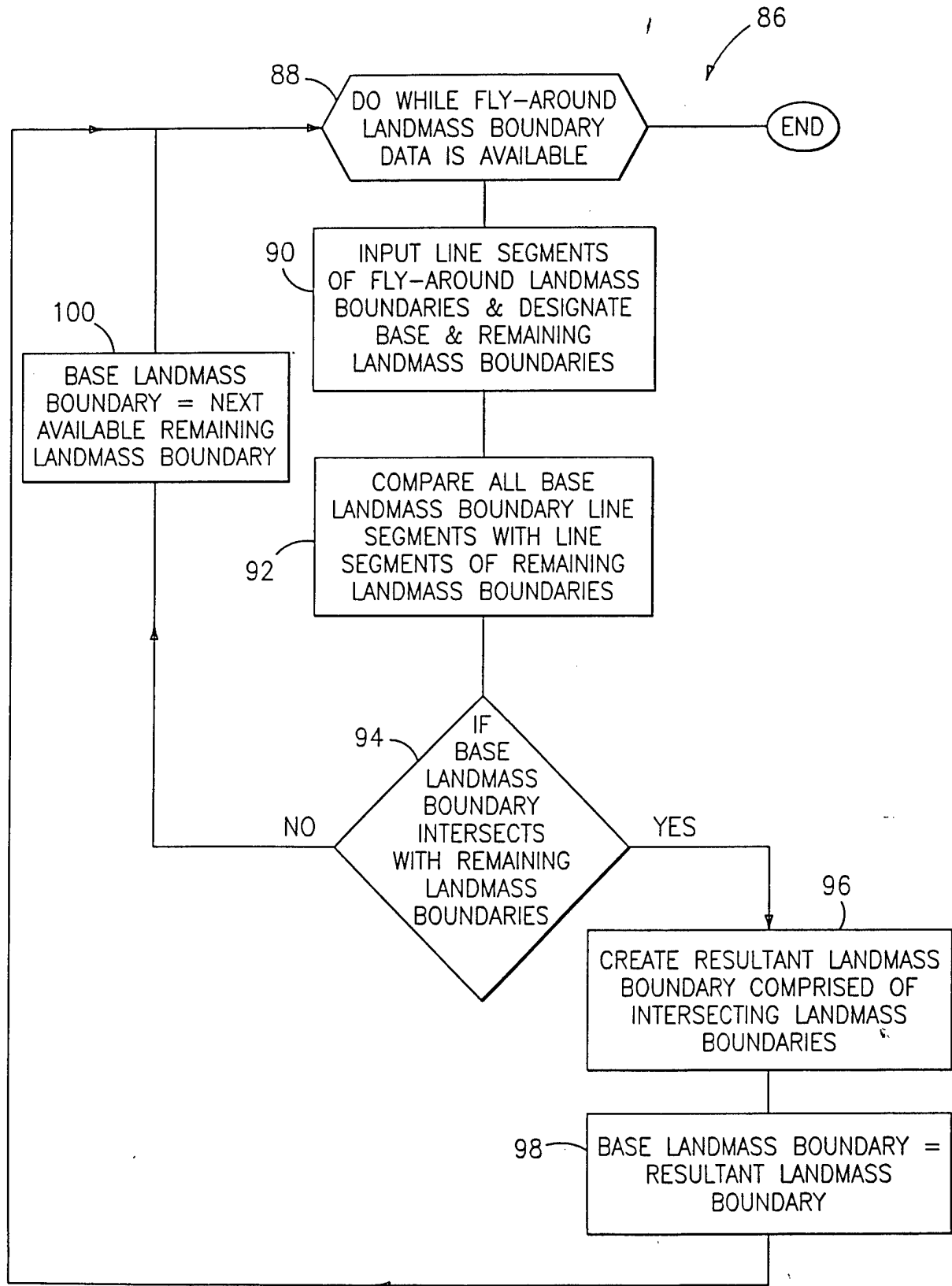


FIG. 11

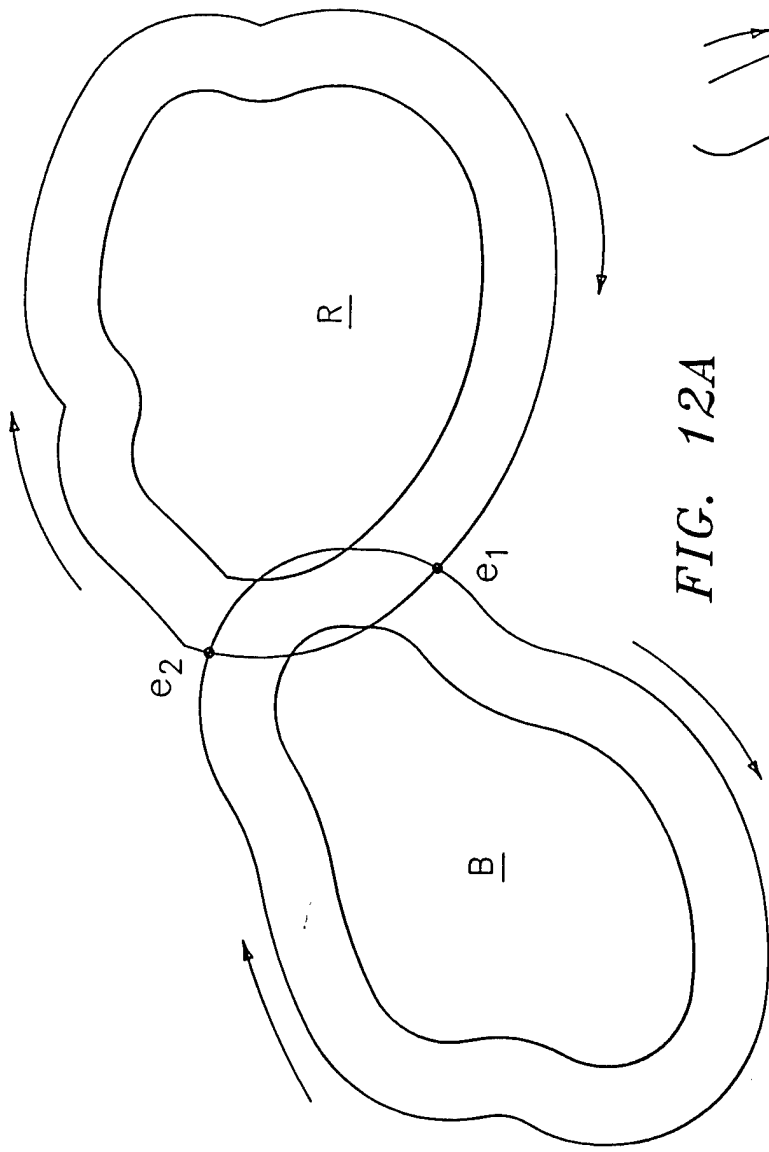


FIG. 12A

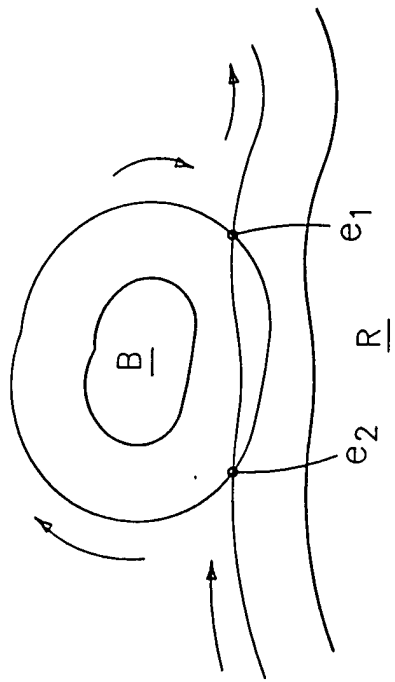


FIG. 12B

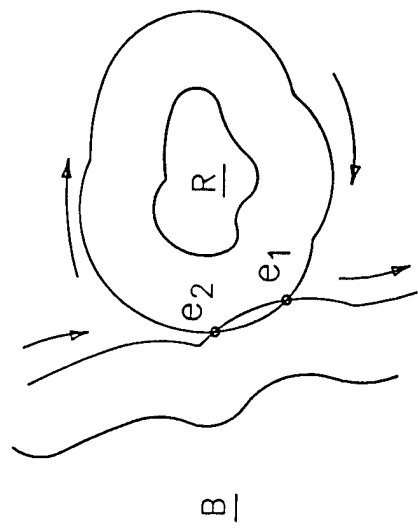


FIG. 12C

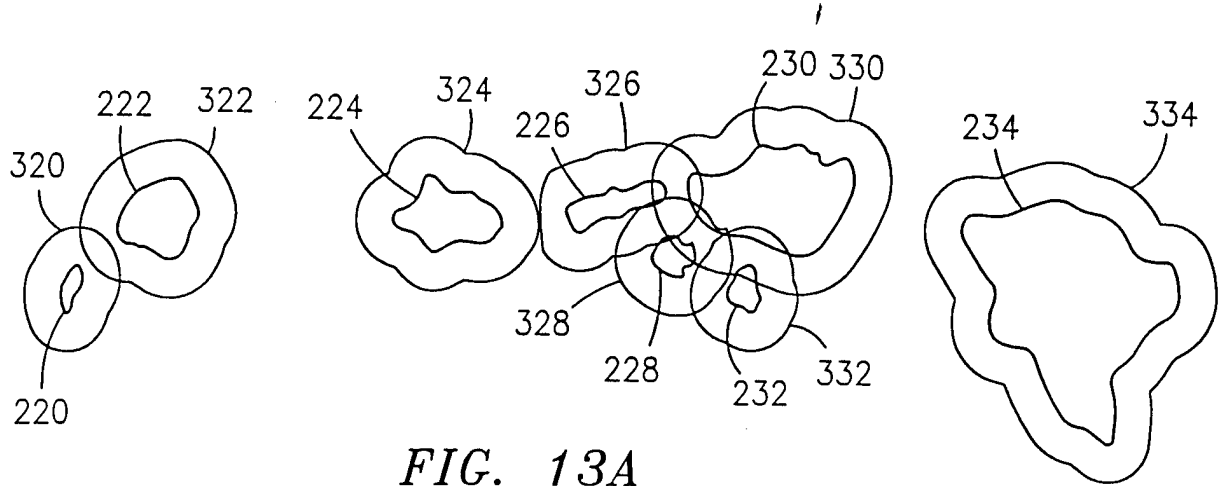


FIG. 13A

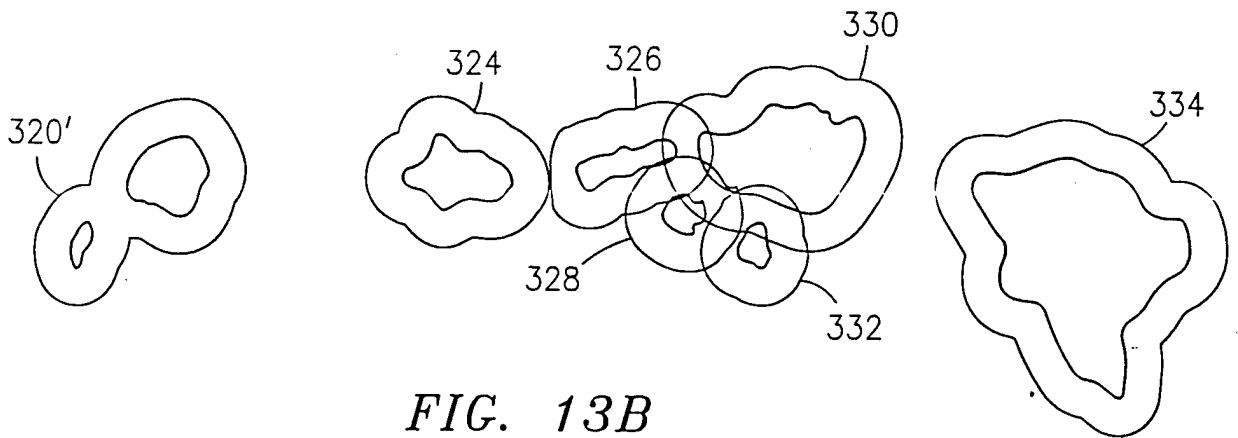


FIG. 13B

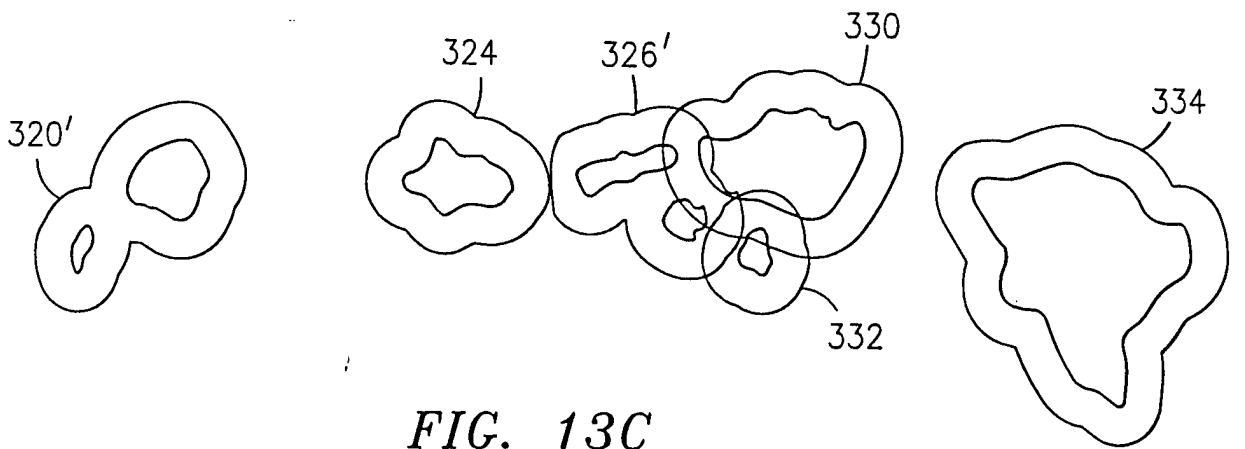


FIG. 13C

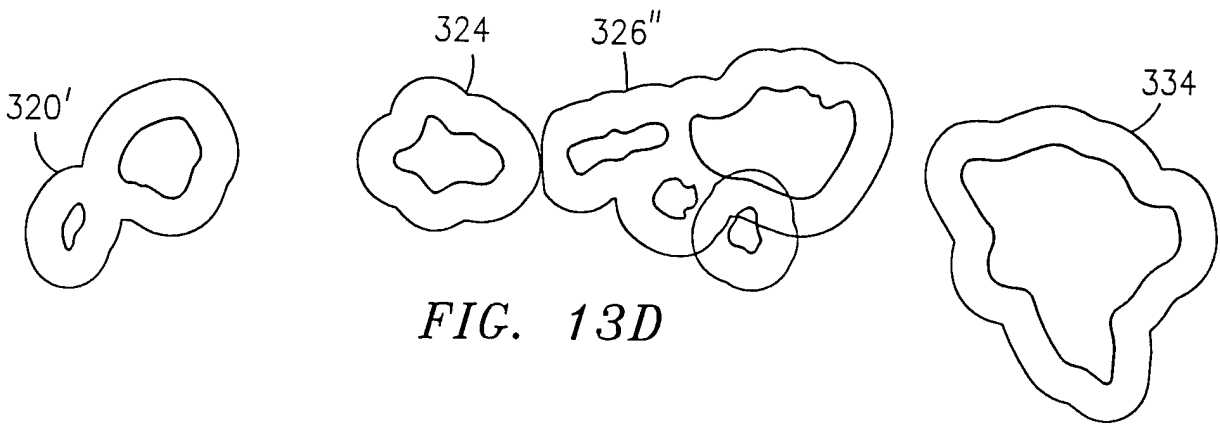


FIG. 13D

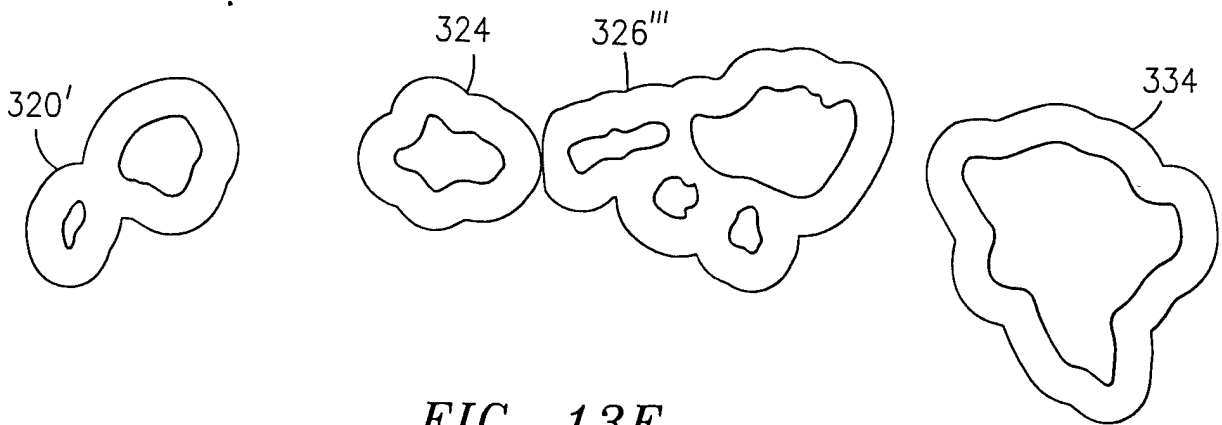


FIG. 13E