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1 Attorney Docket No. 78369

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3 RECONFIGURABLE MULTIPLE COMPONENT LOAD MEASURING DEVICE

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5 STATEMENT OF GOVERNMENT INTEREST

6 The invention described herein may be manufactured and used
7 by or for the Government of the United States of America for
8 governmental purposes without the payment of any royalties
9 thereon or therefor.

10

11 BACKGROUND OF THE INVENTION

12 (1) Field of the Invention

13 The present invention relates to devices and methods for
14 measuring and testing, and more particularly, to devices and
15 methods for measuring forces and force components.

16

17 (2) Description of the Prior Art

18 Existing force measuring devices, force balance and load
19 cells, are complex and expensive systems. Although custom load
20 cells can be fabricated to suit most load measuring requirements,
21 once they have been constructed, their force measurement range
22 and sensitivity are fixed. At the present time, there are no
23 load measurement systems available that can be configured to suit
24 a particular load-measuring requirement without rebuilding the
25 device.

1 U.S. Patent No. 4,094,192 to Watson et al., for example,
2 discloses a six-degree force sensor in which top and bottom
3 sections are coupled to one another by connecting sections.
4 Strain gauges are affixed to these connecting sections.

5 U.S. Patent No. 4,161,874 to Specker et al. discloses a force
6 measuring system in which a plate is supported by a plurality of
7 load cell assemblies. Clevis members pivotally mount each of
8 these load cell assemblies at either end.

9 U.S. Patent No. 4,196,337 to Jewett et al. discloses a torque
10 sensor in which cylindrical rings are spaced apart from one
11 another by elongated beams. Each of these beams has strain
12 gauges mounted thereto.

13 U.S. Patent No. 4,550,617 to Fratgner, et al. discloses a
14 multi-axis force and member transducer, which has a plurality of
15 flexible plates that are orthogonally oriented and interconnected
16 between first and second bodies to measure the forces exerted
17 thereon due to relative movement between the bodies. The
18 interconnection between a first element attached to the first
19 body and a second element attached to the second body is affected
20 by a plurality of first thin flexible plates and a plurality of
21 second thin flexible plates, the ends of the plates being
22 interconnected by a ball-joint coupling. The first thin flexible
23 plates are oriented such that they flex in a direction generally
24 perpendicular to a first axis, while the second thin flexible
25 plates are oriented such that they flex in a direction generally

1 parallel to such axis. Strain gauges may be provided on each of
2 the thin flexible plates to provide an output of the forces.

3 U.S. Patent No. 4,745,565 to Garwin et al. discloses a force
4 sensing type data input device in which two parallel space plates
5 are separated from one another by a plurality of force sensors.

6 U.S. Patent No. 5,063,788 to Ch'Hayder et al. discloses a
7 sensor for measuring three components of force and three
8 components of moment. The sensor is comprised of one unitary
9 mechanical piece comprising two end faces by which may be secured
10 respectively to a body generating force. A central portion
11 comprising six beams with means for measuring deformation are
12 arranged according to a closed triangular architecture between
13 the two end faces.

14 U.S. Patent No. 5,339,697 to Mullin discloses an assembly
15 which measures force along and moments about three orthogonal
16 axes and which reports low cross talk. The structure includes a
17 load cell in which a plurality of links connects couple plates to
18 one another to allow for six degrees of freedom.

19 A need, however, continues to exist for a force measurement
20 device that can be configured for a particular loading range
21 without fabricating new load sensing elements.

22

23 SUMMARY OF THE INVENTION

24 The present invention is a force measurement device that
25 includes a lower support platform. An upper support platform is
26 positioned in spaced relation above the lower support platform.

1 A plurality of members also connects the lower support platform
2 and the upper support platform. On each connecting member there
3 is a load cell, which indicates changes in tension and
4 compression in the members.

5 Also encompassed within the present invention is a method
6 for measuring force. This method comprises the steps of
7 superimposing an upper support platform in spaced relation over a
8 lower support platform; connecting said upper support platform
9 and lower support platform with a plurality of connecting
10 members; applying the force to the upper support platform; and
11 monitoring changes in tension and compression in the members
12 connecting the lower support platform and the upper support
13 platform.

14

15 BRIEF DESCRIPTION OF THE DRAWINGS

16 Other objects, features and advantages of the present
17 invention will become apparent upon reference to the following
18 description of the preferred embodiments and to the drawing,
19 wherein corresponding reference characters indicate corresponding
20 parts in the drawing and wherein:

21 The FIG. is a perspective view of a force measurement device
22 representing a preferred embodiment of the present invention.

23

24 DESCRIPTION OF THE PREFERRED EMBODIMENT

25 Referring to the FIG., the force-measuring device includes a
26 lower platform 10 and an upper platform 12, which is directly

1 superimposed over the lower platform 10 in spaced parallel
2 relation. These upper and lower platforms 10, 12 are designed to
3 be rigid under the design loads of the system, light and fit
4 geometrically into the confined space for the application. They
5 can be any size, distance apart and at any orientation with
6 respect to each other. The exact nature of their relative
7 placement will be determined by the required performance of the
8 device as described in the following paragraphs. The lower
9 platform 10 is connected to the upper platform 12 by six rigid
10 connecting members 14, 16, 18, 20, 22 and 24. These connecting
11 members are fixed to the lower platform respectively by ball
12 joints 26, 28, 30, 32, 34 and 36. The connecting members are
13 respectively connected to the upper platform by upper ball joints
14 38, 40, 42, 44, 46 and 48. The connecting members are
15 constructed with rigid material, which is sufficiently strong to
16 resist deflections and buckling. The ball joints permit only
17 tensile and compressive forces to be transmitted by the
18 connecting members; i.e., the direction of the forces transmitted
19 by the connecting members is solely along the axis of the
20 connection. The ball joints should be fitted to tight tolerances
21 in order to prevent backlash. Poor fits may result in unwanted
22 connecting member pre-loading. Mounted on each of the connecting
23 members there is a load cell 50, 52, 54, 56, 58 and 60
24 respectively. A suitable load cell is commercially available
25 from Transducer Techniques. These load cells produce a voltage
26 output that is proportional to member stresses. The lower

1 platform is secured to a rigid supporting structure and the loads
 2 to be measured (linear forces F and moments M) are applied to a
 3 central point on the upper platform. Under the load, the six
 4 interconnecting members go into tension and compression in a way
 5 which is uniquely determined by the three loading forces, the
 6 three loading moments, the connecting member configuration and
 7 any pre-load on the connecting members. The load cells produce a
 8 voltage output, which is proportional to the member tensions.
 9 The voltages can be recorded to be used later to determine the
 10 applied loads. These voltages are displayed on an
 11 instrumentation package 62.

12 The relationship between connecting member stresses and the
 13 applied load is determined by examining the static force balance
 14 on the device. Each connecting member, i, connects a point
 15 $\mathbf{X}_{1i} = (x_1, y_1, z_1)_i$ on the lower platform to a point on the upper
 16 platform $\mathbf{X}_{2i} = (x_2, y_2, z_2)_i$. Because the members transmit forces along
 17 their axes (defined as the vector from \mathbf{X}_{1i} to \mathbf{X}_{2i} , \mathbf{v}_i) with a
 18 magnitude equal to the tension in the member, the net force
 19 imposed on the upper platform by a single connection member is
 20 the projection of the tension vector into the three component
 21 directions.

$$22 \quad (F_x, F_y, F_z)_i = T_i \left(\frac{\mathbf{v}_i}{\|\mathbf{v}_i\|} \right) \quad (1)$$

23 where

24 F_x is the force in the x direction

25 F_y is the force in the y direction

1 F_z is the force in the z direction
2 T_i are the tensions in the connecting members ($i=1$ to 6).
3 The net moment imposed on the upper platform by a single
4 connection member about a reference point (the point of load
5 application) is the tension vector crossed into the distance d_i
6 between the connection point \mathbf{x}_{2i} and the reference point
7 $\mathbf{x}_r = (x_r, y_r, z_r)$, $\mathbf{d}_i = \mathbf{x}_{2i} - \mathbf{x}_r$.

$$8 \quad (M_x, M_y, M_z)_i = T_i \left(\frac{\mathbf{v}_i}{\|\mathbf{v}_i\|} \right) \times (d_i) \quad (2)$$

9 where

10 M_x is the moment about the x axis
11 M_y is the moment about the y axis
12 M_z is the moment about the z axis.

13
14 The total force on the upper plate is a sum of the contributions
15 by all six (6) connection members. Equations (1) and (2) are
16 expanded and rewritten as

$$\begin{aligned} 17 \quad F_x &= (A_{11})T_1 + (A_{21})T_2 + (A_{31})T_3 + (A_{41})T_4 + (A_{51})T_5 + (A_{61})T_6 \\ 18 \quad F_y &= (A_{12})T_1 + (A_{22})T_2 + (A_{32})T_3 + (A_{42})T_4 + (A_{52})T_5 + (A_{62})T_6 \\ 19 \quad F_z &= (A_{13})T_1 + (A_{23})T_2 + (A_{33})T_3 + (A_{43})T_4 + (A_{53})T_5 + (A_{63})T_6 \\ 20 \quad M_x &= (A_{14})T_1 + (A_{24})T_2 + (A_{34})T_3 + (A_{44})T_4 + (A_{54})T_5 + (A_{64})T_6 \\ 21 \quad M_y &= (A_{15})T_1 + (A_{25})T_2 + (A_{35})T_3 + (A_{45})T_4 + (A_{55})T_5 + (A_{65})T_6 \\ 22 \quad M_z &= (A_{16})T_1 + (A_{26})T_2 + (A_{36})T_3 + (A_{46})T_4 + (A_{56})T_5 + (A_{66})T_6 \end{aligned} \quad (3)$$

23

1 where A_{ij} are coefficients ($i=1$ to 6 and $j=1$ to 6) determined from
2 the geometry of the interconnections.

3 The optimal platforms and connecting member configuration
4 and pre-loads for a specific application must conform to two
5 major constraints. First, to guarantee that there will not be
6 any sensor overloads, the stress in individual members must not
7 exceed the maximum rated values for sensor loads within the
8 range:

9
$$F_{x\min} < F_x < F_{x\max}, F_{y\min} < F_y < F_{y\max}, F_{z\min} < F_z < F_{z\max}, \quad (4)$$

10

11
$$M_{x\min} < M_x < M_{x\max}, M_{y\min} < M_y < M_{y\max}, M_{z\min} < M_z < M_{z\max}. \quad (5)$$

12

13 In addition, the stresses in the members does not exceed the
14 maximum rated values when the assembly is under no load (tensions
15 resulting from pre-loading). Second, to maximize the device
16 sensitivity, at some operating point within the loading range,
17 the stress in each connecting member reaches its maximum rated
18 value. Similarly, at some operating point within the loading
19 range, the stress in each connecting member reaches its minimum
20 rated value.

21 Although it may be possible to analytically invert the
22 system of equations (1) through (3) and solve for optimal
23 connector configurations, such an approach has not been
24 implemented. Instead, a trial and error approach has been used.
25 The connector locations on the lower platform x_{1i} are fixed, a

1 reasonable configuration is assumed (x_{2i}, x_r are selected) and the
2 load relationship matrix A is computed for the geometry. To
3 evaluate the geometry, the connector tensions are computed using
4 the inverse of equation (3) for many load combinations within the
5 desired load range. If the constraints outlined in the preceding
6 section are met, the geometry is accepted. If they are not, new
7 geometries are formulated and tested until a suitable
8 configuration is found.

9 The system stiffness is needed in order to determine the
10 natural frequency of the sensor assembly. It is assumed that the
11 upper supporting platform is rigid and the motion of a point on
12 the upper platform is algebraically related to the displacement
13 and rotation of the platform, according to the formula:

$$14 \quad dx_{2i} = f_i(dx, \alpha, \beta, \gamma) \quad (6)$$

15 where

16 dx_{2i} is the displacement of point x_{2i}

17 dx is the displacement of the upper platform

18 α is the rotation of the upper platform with respect
19 to the x axis

20 β is the rotation of the upper platform with respect
21 to the y axis

22 γ is the rotation of the upper platform with respect
23 to the z axis.

24

25 The change in length of a connecting member as a result of
26 the motion of the upper platform is

$$dl_i = \|x_{2i} - x_{1i}\| - \|x_{2i} + dx_{2i} - x_{1i}\|. \quad (7)$$

2

3 Because each load-sensing element in each connecting member
4 acts as a spring with stiffness Kc , and the tension in each
5 member is proportional to its change in length

$$T_i = (Kc)(dl_i) \quad (8)$$

7

8 The platform loads associated with these tensions are
9 determined using Equation (1). A stiffness matrix is thus
10 computed having the form

$$[F_x, F_y, F_z, M_x, M_y, M_z] = [dx, dy, dz, \alpha, \beta, \gamma][K_{ij}]. \quad (9)$$

12

13 If the masses and moments of inertia of the sensor system
14 and the connected test hardware are known, the natural
15 frequencies of the system can be determined. If these
16 frequencies are not acceptable, the configuration can be changed
17 and the system reevaluated.

18 To miniaturize the current design, the connecting members
19 and load cells can be replaced by miniaturized connecting
20 members, which are instrumented with strain gauges to serve as
21 tension compression load cells. The space savings realized by
22 constructing custom miniaturized load cells must be weighed
23 against the expense of constructing these connecting members.

24 It will be appreciated that an advantage of this invention
25 is that the assembly can be configured for a particular loading
26 range without fabricating new load sensing elements. The modular

1 construction of the present design not only permits custom
2 configurations of the load sensing system, but also permits the
3 load sensing elements to be individually removed and serviced
4 when required. This modularity enhances the device capability
5 and reduces its overall cost to construct and maintain.

6 While the present invention has been described in connection
7 with the preferred embodiments, it is to be understood that other
8 similar embodiments may be used or modifications and additions
9 may be made to the described embodiment for performing the same
10 function of the present invention without deviating therefrom.
11 Therefore, the present invention should not be limited to any
12 single embodiment, but rather construed in breadth and scope,
13

1 Attorney Docket No. 78369

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3 RECONFIGURABLE MULTIPLE COMPONENT LOAD MEASURING DEVICE

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5 ABSTRACT OF THE DISCLOSURE

6 A force measurement device that includes a lower support
7 platform. An upper support platform is positioned in spaced
8 relation above the lower support platform. A plurality of
9 members also connects the lower support platform and the upper
10 support platform. On each connecting member there is a load
11 cell, which indicates changes in stress in the members.

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