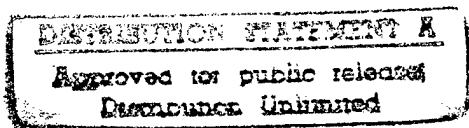


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NOTICE

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2  
3 FUEL DELIVERY SYSTEM

4  
5 STATEMENT OF GOVERNMENT INTEREST

6 The invention described herein may be manufactured and used  
7 by or for the Government of the United States of America for  
8 governmental purposes without the payment of any royalties  
9 thereon or therefor.

10  
11 BACKGROUND OF THE INVENTION

12 (1) Field of the Invention

13 The present invention generally relates to a fuel delivery  
14 system which eliminates the risk of a catastrophic fuel  
15 explosion.

16 (2) Description of the Prior Art

17 Torpedoes are routinely subjected to land based testing to  
18 evaluate the on board propulsion system. In such land based  
19 testing, one system involves situating and immobilizing a torpedo  
20 in a test cell. To simulate load on the drive shaft of the  
21 propulsion system of the torpedo, the propeller drive shaft is  
22 mechanically connected to a torque device, e.g., a dynamometer.  
23 The test cell is an air-tight and water-tight structure. The  
24 test cell is flooded with water such that an immobilized torpedo  
25 therein is completely submerged in water during the test. The  
26 water in the test cell is controllably pressurized to duplicate a

1 depth condition of interest. The torpedo's propulsion system is  
2 then tested by running its on board motor. The engine can be run  
3 at different speeds and under different tank pressures to  
4 comprehensively assess the performance capabilities of the  
5 torpedo's propulsion system under a wide variety of simulated  
6 operating conditions.

7 A heavy weight torpedo, such as those tested in this manner,  
8 is an external combustion engine powered by Otto fuel. The Otto  
9 fuel used to power the engine is a non-corrosive liquid fuel  
10 monopropellant developed specifically for use in underwater  
11 propulsion systems. "Otto fuel", for purposes of this  
12 application, encompasses "Otto Fuel II", which is a known,  
13 combustible torpedo fuel based on propylene glycol dinitrate as a  
14 propellant. Otto Fuel II also typically contains smaller amounts  
15 of adjuvants such as a stabilizer or desensitizer (e.g., 2-  
16 nitrodiphenylamine), and a plasticizer (e.g., di-n-butyl  
17 sebacate). Pressurized Otto fuel, in general, is less stable and  
18 more susceptible to inadvertent explosion. In particular, Otto  
19 Fuel II can explode or deflagrate if it is confined and subjected  
20 to pressures in excess of 350 psi, or if it is atomized at  
21 pressures in excess of 100 psi.

22 In order to run the propulsion system on board a test  
23 torpedo for an extended period of time without interruption of  
24 the test sequence, a relatively large quantity of Otto fuel must  
25 be made available to the test torpedo. In prior land based  
26 testing of heavy weight torpedoes, for instance, in excess of 100

1 gallons of pressurized Otto fuel has been stored on board the  
2 test torpedo. If this quantity of pressurized, combustible Otto  
3 fuel inadvertently detonates within the test cell, the fuel  
4 explosion emanating from the fuel stored aboard the test torpedo  
5 can cause serious structural damage. Even if a test cell largely  
6 contains and absorbs the blast to protect the surrounding area,  
7 the explosion can result in the loss of test facility assets at  
8 least until any necessary repairs are made to the test cell  
9 leading to costly program delays.

10 A system for transporting and handling fluids under high  
11 pressure is generally described in United State Patent No.  
12 4,446,804 to Kristiansen et al. which discloses a method of  
13 transporting oil and gas under high pressure in tanks on board a  
14 ship. The transportation of oil and gas under high pressure in  
15 tanks on board a ship is carried out by loading and unloading the  
16 oil/gas utilizing a suitable pressurized liquid (e.g., water) in  
17 the individual tanks, whereby during loading, a tank or a group  
18 of tanks containing pressurized liquid are filled with oil and  
19 gas while the pressurized liquid simultaneously is displaced into  
20 the next tank or group of tanks which are to be filled, after  
21 which the next tank/group of tanks are filled and the pressurized  
22 liquid displaced into a third tank/group of tanks, and so forth.  
23 During unloading, the cargo in one tank or one group of tanks is  
24 removed by introducing a pressurized liquid into the tank/group  
25 of tanks, and unloading of the cargo in the next tank/group of

1 tanks occurs by transferring the pressurized liquid from the  
2 first tank/group of tanks to the next.

3 Also, United States Patent No. 3,874,399 to Ishihara  
4 discloses an oil delivery system for high melting point oils in a  
5 tank. The oil delivery system effects the discharge of  
6 solidified or semi-solidified oil remaining in a tank after  
7 primary delivery of the bulk of the oil therefrom by insertion of  
8 a nozzle in the free space within the tank, and oil of the same  
9 kind as the oil in the primary delivery is heated and directed  
10 through the nozzle onto the residual oil, which is thereby  
11 melted, and rendered easily movable.

12 None of the above prior art systems address and solve the  
13 problems raised and risks posed by exposure of relatively large  
14 quantities of a combustible engine fuel to pressure.

#### 15 SUMMARY OF THE INVENTION

16 It is therefore an object of the present invention to  
17 provide a fuel delivery system for a test cell which eliminates  
18 the risk of a catastrophic fuel explosion during testing of  
19 vehicles, such as torpedoes.  
20

21 In order to accomplish this and other objects of the  
22 invention, the present invention provides a fuel delivery system,  
23 which effectively limits the quantity of fuel under  
24 pressurization at any given time, such as during a test sequence  
25 for a vehicle engine. The present invention thus effectively  
26 reduces the magnitude of an explosion, and thus the scale of any

1 damage, associated with any inadvertent detonation of pressurized  
2 fuel. In one embodiment of the invention, the inventive fuel  
3 delivery system supports a test cell used in land-based testing  
4 of an underwater vehicle (e.g., a torpedo).

5 The fuel delivery system of this invention includes a  
6 primary fuel storage tank that is mechanically isolated from and  
7 located outside the test cell, such as where the propulsion  
8 system of underwater vehicle is actually tested, via an  
9 intervening fuel support cell. The fuel support cell is also  
10 located outside the test cell and houses means which controllably  
11 limit the quantity of fuel, as supplied by the primary storage  
12 tank, that is pressurized and available at any given instant to  
13 support combustion requirements of an engine being tested. The  
14 fuel support cell houses both the test vessel in which the  
15 vehicle is actually tested, and an external fuel delivery system  
16 used to pressurize, store, and feed controlled and reduced  
17 quantities of pressurized fuel to the test vehicle. The present  
18 invention thus effectively reduces the magnitude of an explosion,  
19 and thus the scale of any damage associated with any inadvertent  
20 detonation of pressurized fuel.

21 The external fuel delivery system is "external" in the sense  
22 of being located outside the test cell where the vehicle is  
23 tested. More particularly, the external fuel delivery system  
24 includes a pumping system, an intermediate fuel storage tank, and  
25 an arrangement of fuel lines adequate to permit fluid  
26 communication between these components. Fuel is drawn from the

1 primary fuel storage tank located outside the fuel support cell  
2 and fed into the fuel support cell for handling (processing) by  
3 the external fuel delivery system. Before being fed into the  
4 test cell for combustion in the engine aboard the test vehicle,  
5 fuel is first pressurized through a pumping system and then  
6 temporarily stored in small increments in the intermediate fuel  
7 storage tank, e.g., about 5 gallons at a time in the case of Otto  
8 fuel.

9 In one preferred arrangement, the multi-stage pumping system  
10 involves a two-stage operation. In the first stage, a boost pump  
11 creates adequate head pressure in the fuel to prevent cavitation  
12 of the fuel in a second succeeding pump stage. Fluid pressure is  
13 increased in the second stage to ensure an uninterrupted supply  
14 of fuel is maintained to the intermediate fuel storage tank.  
15 The fuel is pumped against a reference pressure established in  
16 the intermediate storage tank which is referenced to the depth  
17 pressure condition concurrently established in the test cell for  
18 testing of underwater vehicles. To accomplish this, a pressure-  
19 over-liquid tank arrangement is employed to create a back  
20 pressure such that the intermediate tank can both temporarily  
21 store small quantities of pressurized fuel (obtained from the  
22 two-stage pumping system) and transmit the fuel, as needed, into  
23 the test cell to power the combustion engine aboard the test  
24 vehicle, such as a torpedo. The pumped fuel temporarily stored  
25 within the pressure-over-liquid tank is separated from the air in  
26 the upper regions of the vessel by an intervening blanket of

1 water to preclude air entrapment in the fuel. A control means is  
2 also joined to the primary fuel storage tank, the pumping system,  
3 and the intermediate fuel storage tank, which controllably limits  
4 the quantity of fuel provided to the test cell.

5 By virtue of the external fuel delivery system of the  
6 present invention, it is possible to modify the fuel tank  
7 provided aboard the vehicle, e.g., a torpedo, such that it need  
8 only store extremely small quantities of fuel directly aboard the  
9 vehicle within the test cell. Namely, only a minimal amount of  
10 fuel necessary to power the engine and tolerate instantaneous  
11 fuel flow fluctuations in the fuel lines need be stored directly  
12 aboard the vehicle in the test cell. In one embodiment of the  
13 invention, only about 0.5 gallons of Otto fuel are stored  
14 directly aboard a test torpedo, instead of 100 gallons or more of  
15 Otto fuel as encountered in practice preceding this invention.  
16 Therefore, the force of any explosion associated with the fuel  
17 stored aboard the torpedo alone is significantly mitigated.

18 In a further embodiment, the external fuel delivery system  
19 is also supplemented with blast suppression fitting means (e.g.,  
20 detonation traps) at the inlet and outlet of the external fuel  
21 delivery system. In this embodiment, a detonation trap is  
22 provided in the portion of the fuel line that feeds the external  
23 fuel delivery system from the primary fuel storage tank at a  
24 location inside the fuel support cell, and another detonation  
25 trap is provided in the fuel line at a location inside the fuel  
26 support cell where fuel has cleared the pumping and storage



1 components of the external fuel delivery system but has not yet  
2 entered the test cell. The detonation traps help confine any  
3 pressurized fuel explosion within the fuel support cell or test  
4 cell preventing propagation of a fuel blast through fuel lines.  
5 For example, this arrangement prevents any detonation of the  
6 pressurized fuel from propagating into the fuel support cell from  
7 the test cell, or vice versa, or from propagating from the fuel  
8 support or test cell back into the primary reservoir of stored  
9 fuel situated outside the fuel support cell. The detonation  
10 traps can also be used in conjunction with decomposition  
11 arrestors which are utilized to quench any deflagration that  
12 could manifest in the fuel line.

13 Therefore, the fuel delivery system of the present invention  
14 is highly advantageous from cost and safety standpoints as it  
15 effectively eliminates the risk of a catastrophic fuel explosion  
16 during testing of vehicles such as torpedoes.

#### 17 18 BRIEF DESCRIPTION OF THE DRAWING

19 A more complete understanding of the invention and many of  
20 the attendant advantages thereto will be readily appreciated as  
21 the same becomes better understood by reference to the following  
22 detailed description when considered in conjunction with the  
23 accompanying drawing wherein:

24 The Figure illustrates a flow diagram showing a fuel  
25 delivery system in accordance with the invention.



1 to a torque device of a conventional kind, e.g., a dynamometer  
2 (not shown). During a test run, the test cell 25 is completely  
3 flooded and filled with water such that the immobilized torpedo  
4 26 therein is completely submerged in water during the test.  
5 The water in the test cell 25 is controllably pressurized by a  
6 depth control of conventional design and usage to duplicate the  
7 depth condition of interest. The torpedo's propulsion system is  
8 then tested by running its on board motor.

9 The components of the propulsion system for torpedo 26,  
10 other than its on board fuel storage tank 27, include components  
11 common to conventional torpedo propeller systems which will be  
12 appreciated by those of ordinary skill in the art and which do  
13 not form essential parts of the present invention of interest.  
14 For example, suitable conventional torpedo propulsion systems  
15 include a propeller mounted at the stern end of an internally  
16 mounted propeller drive shaft, and the propeller is driven  
17 rotationally thereby. The drive shaft is driven by a combustion  
18 engine powered by Otto fuel stored aboard the torpedo.

19 During a test run, the torpedo's propulsion system is tested  
20 under varying conditions of pressure and speed as conducted in  
21 test cell 25. The fuel needed to power the torpedo's propulsion  
22 system directly draws upon a small reservoir of Otto fuel, e.g.,  
23 only about 0.5 gallons of Otto Fuel II, stored in tank 27 aboard  
24 the torpedo itself. By modifying the on board fuel tank 27 to  
25 only store the minimal amount of fuel necessary to power the  
26 engine and tolerate instantaneous fuel flow fluctuations in the

1 fuel lines, the explosive force of any inadvertent explosion  
2 associated with the fuel stored aboard the torpedo 26 alone is  
3 significantly reduced. In order to replenish the fuel in fuel  
4 tank 27, as it is consumed to power the torpedo's engine, the  
5 external fuel delivery system 10 of the present invention is  
6 employed.

7 The external fuel delivery system 10 is supplied fuel from  
8 the primary fuel storage tank 15 located outside the fuel support  
9 cell 12. Tank 15 has a float switch 9C to prevent complete  
10 evacuation of the tank if the fuel level reaches a lower  
11 threshold level. Flow of fuel from the primary fuel storage tank  
12 15 to fuel support cell 12 is controlled by speed control valve  
13 13, which is provided with adjustable positions. Flow meter 8D  
14 permits the feed rate to be monitored to control the setting of  
15 the speed control valve 13. A blanket of water 28A floats on top  
16 of the fuel 16A stored in the primary storage tank 15 to negate  
17 the possibility of air entrapment in the fuel 16A. Fuel that has  
18 been drawn from the primary fuel storage tank 15 outside the fuel  
19 support cell and fed into the fuel support cell 12 for handling  
20 by the external fuel delivery system 10 is first pressurized  
21 through a pumping system 20, and then temporarily stored in small  
22 increments in the intermediate fuel storage tank 17, e.g., about  
23 5 gallons at a time in the case of Otto Fuel II, before being fed  
24 into the test cell 25 for combustion in the engine aboard the  
25 test vehicle 26.

1           In greater detail, as shown in the Figure, the pumping  
2 system 20 is a two-stage pumping system, and its purpose is to  
3 pump fuel to intermediate fuel storage tank 17. The first stage  
4 of the pumping system 20 is a centrifugal boost pump 4 that  
5 creates a head pressure of 30 psi in the fuel. The second stage  
6 of the pumping system 20 is a positive displacement pump 1 that  
7 requires a 30 psi head, provided by the boost pump 4. The boost  
8 pump 4 is used because Otto fuel, when allowed to cavitate and  
9 then compress, can cause an adiabatic compression of the fuel  
10 leading to a pump explosion. The boost pump 4 is driven by its  
11 motor 14B and positive displacement pump 1 by its respective  
12 motor 14A. Pressure switch 19A monitors the head pressure  
13 created by the boost pump 4 and it will shut off the motor 14A of  
14 the positive displacement pump 1 if minimal head pressure has not  
15 been established by the boost pump 4. The positive displacement  
16 pump 1 can create fuel pressures of up to about 2,000 psi.  
17 Relief valves 21A and 21B, used in conjunction with flow meter 8C  
18 and one-way control valve 5A, are provided in support of pumping  
19 system 20. The relief valves 21A and 21B allow the fuel lines 11  
20 to be bled off if excessive pressures are created by either the  
21 boost pump 4 or positive displacement pump 1. If fuel is bled  
22 off in this manner, one-way control valve 5A allows the fuel to  
23 be recycled back into the primary storage tank 15. Flow control  
24 valve 23 is provided for use in conjunction with speed control  
25 valve 13 to adjust the amount of fuel delivered from pumps 4 and  
26 1.

1           The fuel pumped by the positive displacement pump 1 is  
2 pumped against a depth control system reference pressure through  
3 the use of a pressure-over-liquid tank arrangement established in  
4 intermediate fuel storage tank 17. The "depth control system  
5 reference pressure" created in the air blanket 2B above the water  
6 blanket 28B in intermediate tank 17 is established by an air  
7 supply feed system 3. The air supply feed system 3 feeds from a  
8 separate air supply, which is shared by the water in the pressure  
9 vessel of test cell 25. During a test run, solenoid valve 22B  
10 (normally closed) is energized and opened, and solenoid valve 22C  
11 (normally open) closed, so that the air supply system 3 can  
12 create a back pressure in the intermediate tank 17 which will be  
13 the same as the water pressure being created in the pressure  
14 vessel shell of the test cell 25. The fuel 16B within the  
15 intermediate tank 17 is separated from the pressurized air 2B by  
16 an intervening blanket of water 28B to negate the possibility of  
17 air entrapment in the fuel 16B in intermediate tank 17. Once the  
18 test run is completed, solenoid valve 22B is de-activated and  
19 solenoid valve 22C re-opened to vent the pressurized air in  
20 intermediate storage tank 17.

21           The fuel level within intermediate storage tank 17 is  
22 controlled by float switches 9A and 9D. When float switch 9A, as  
23 a pumping switch, is energized (activated) by the fluid level  
24 falling below a given amount, it will activate the pumps in the  
25 pumping system 20 to keep re-filling the tank 17 with fuel. One-  
26 way control valve 5B is opened when the boost pump 4 and positive

1 displacement pump 1 are running. Flow meter 8B, pressure sensor  
2 18A, and temperature sensor 24A are also used in support of the  
3 pumping operation to intermediate tank 17. When float switch 9D  
4 is activated by the fuel level falling to another lower threshold  
5 level, it will shut off fuel flow by closing solenoid valve 22A  
6 to prevent air from being pumped into the test vehicle 26 and  
7 test cell 25. Float switch 9B is a safety switch which shuts off  
8 the pumping system 20 if the fuel level becomes too high in  
9 intermediate fuel storage tank 17.

10 By these means, the intermediate tank 17 can temporarily  
11 store relatively small quantities of pressurized fuel (obtained  
12 from the two-stage pumping system 20) before its transport, as  
13 needed, into the test cell 25 to power the combustion engine  
14 aboard the test torpedo 26.

15 During the test run, the fuel temporarily stored in  
16 intermediate fuel storage tank 17 is fed to the fuel tank 27  
17 aboard test vehicle 26 by energizing and opening solenoid valve  
18 22A (normally closed), where the fuel is monitored in transit by  
19 pressure sensor 18B, temperature sensor 24B, and flow meter 8A.  
20 The back pressure created by the pressurized air 2B formed in  
21 intermediate fuel storage tank 17 forces the fuel out of the tank  
22 17 and through the intervening fuel lines to fuel tank 27 aboard  
23 test torpedo 26. Pressure switch 19B operates to close solenoid  
24 valve 22A and concurrently open solenoid valve 22D (which  
25 supports another recycle fuel line), if an insufficient fuel line  
26 pressure is detected in the fuel line exiting intermediate tank

1 17. The fuel stored aboard the test torpedo 26 is fed to a drive  
2 system (not shown) through flow meter 8E during the test run.  
3 The external fuel delivery system 10, including the pumping  
4 system 20 and intermediate fuel storage tank 17, are all  
5 supported by an arrangement of fuel lines 11 adequate to permit  
6 fluid communication to-and-from and between these components.  
7 Further, the fuel delivery system 10 is monitored and controlled  
8 remotely through the series of solenoid valves, pressure  
9 switches, pressure and temperature sensors and flow meters, as  
10 discussed herein and shown in the Figure. A computer controller  
11 6 is provided in communication with the components of this device  
12 to shut down the pumping system 20 or issue a warning if an error  
13 occurs. The computer controller 6 is linked to receive  
14 information from float switch 9C in primary storage tank 15 and  
15 float switches 9A, 9B, and 9D in the intermediate storage tank  
16 17. Controller 6 can also be joined to flow meters 8A, 8B, 8C,  
17 8D and 8E to monitor fuel flow through the system. Likewise,  
18 computer controller 6 is joined to pressure sensors 18A and 18B  
19 and temperature sensors 24A and 24B to detect the pressure and  
20 temperature of the pressurized fuel. Computer controller 6  
21 provides a control signal to pump motors 14A and 14B and solenoid  
22 valves 22A, 22B, 22C, and 22D to enable controller 6 to shut down  
23 fuel flow to the test cell 25 in circumstances that indicate an  
24 error.

25 By virtue of the external fuel delivery system 10 of the  
26 present invention, it is possible to modify the on board fuel



1 tank 27 provided aboard the torpedo 26, such that it need only  
2 store extremely small quantities of fuel directly aboard the test  
3 vehicle (located within the test cell 25). Namely, only a  
4 minimal amount of fuel necessary to power the engine and tolerate  
5 instantaneous fuel flow fluctuations in the fuel lines need be  
6 stored directly aboard the test vehicle 26 in the test cell 25.  
7 In one embodiment of the invention, only about 0.5 gallons of  
8 Otto fuel are stored directly aboard a test torpedo 26, instead  
9 of 100 gallons or more of Otto fuel as encountered in practice  
10 preceding this invention. Therefore, the force of any explosion  
11 associated with the fuel stored aboard the torpedo alone is  
12 significantly mitigated. Despite the significantly reduced  
13 amount of fuel stored aboard the test torpedo as made possible by  
14 the present invention, it has been found that the size of the on  
15 board tank 27 and the flow characteristics of the pumping system  
16 20 are nonetheless adequate to support any instantaneous fuel  
17 flow fluctuations.

18 In a further embodiment, the external fuel delivery system  
19 10 is also supplemented with blast suppression fitting means  
20 (e.g., detonation traps) of a conventional design, at the inlet  
21 and outlet of the external fuel delivery system 10. In this  
22 embodiment, the detonation trap 7A is provided in the portion of  
23 the fuel line that feeds the external fuel delivery system 10  
24 from the primary fuel storage tank 15 at a location inside the  
25 fuel support cell 12. Another detonation trap 7B is provided in  
26 the fuel line at a location inside the fuel support cell 12 where

1 fuel has cleared the pumping and storage components of the  
2 external fuel delivery system 10 but has not yet entered the test  
3 cell 25. A detonation trap 7C is also included as shown in the  
4 Figure. This provision of detonation traps helps to confine any  
5 explosion of pressurized fuel to within the fuel support cell or  
6 test cell, where it occurs, respectively, by preventing  
7 propagation of such fuel blasts through the fuel lines into  
8 adjacent cells otherwise in fluid communication with the cell in  
9 which the blast occurs and/or the area outside and surrounding  
10 the particular cell in which the blast occurs. For example, this  
11 arrangement prevents any detonation of the pressurized fuel from  
12 propagating into the fuel support cell 12 from the test cell 25,  
13 or vice versa, or from propagating from the fuel support 12 or  
14 test cell 25 back into the primary reservoir of stored fuel 15  
15 situated outside the fuel support cell 12. The detonation traps  
16 can also be used in conjunction with decomposition arrestors (not  
17 shown) of conventional nature which are utilized to quench any  
18 deflagration that could manifest in the fuel line.

19 The subject invention is capable of delivering fuel to  
20 support all required heavy weight torpedo engine fuel flow and  
21 pressure requirements while exposing less than about 5½ gallons  
22 of Otto Fuel to pressure at any given instance during the test  
23 sequence. It has been determined that an explosion of about  
24 5 to 6 gallons or less of Otto Fuel will not result in serious  
25 damage to a test facility. This advantage is realized with

1 minimal impact upon the conduct of the test or the validity of  
2 the test results.

3 Although not particularly limited, the type of combustion  
4 fuel employed to power the engine on board the torpedo optimally  
5 is a so-called Otto fuel known to those of skill in the art. A  
6 suitable formulation of "Otto fuel", for purposes of this  
7 invention, is an "Otto Fuel II" formulation involving a mixture  
8 containing propylene glycol dinitrate as a propellant material,  
9 an amount of di-n-butyl sebacate effective to plasticize the  
10 mixture, and an amount of 2-nitrophenylamine effective to  
11 stabilize the mixture. The propellant, i.e., propylene glycol  
12 dinitrate, is the predominant component of the mixture. One  
13 exemplary formulation of the Otto Fuel II suitable for use  
14 practicing in the invention includes, by weight percentages, the  
15 following:

16 (a) about 76% propylene glycol dinitrate ( $C_3H_6N_2O_6$ ;

17 CAS No.: 6423-43-4);

18 (b) about 22.5% di-n-butyl sebacate ( $C_4H_9OCO(CH_2)_8OCOC_4H_9$ ); and

19  
20 (c) about 1.5% 2-nitrophenylamine ( $o-C_6H_5NHC_6H_4NO_2$ );

21 where the maximum moisture ( $H_2O$ ) content of the mixture does  
22 not exceed 0.10% by weight. This formulation of Otto Fuel II has  
23 the following physical characteristics (at 77°F):

24 density: 1.232 g/ml

25 freezing point: -18.4°F

26 vapor pressure: 0.0877 mm Hg

1 viscosity 4.4 cp  
2 surface tension: 34.45 dynes/cm  
3 flash point: 265°F  
4 water satur. pt.: 0.31%  
5 heat capacity: 0.445 BTU/lb°F  
6 solubilities: insoluble: water, ethylene glycol  
7 very soluble: alcohols, gasoline, acetone  
8

9 In view of the foregoing, it is seen that the invention  
10 provides an effective solution to previous problems related to  
11 risks of catastrophic fuel explosion during land based testing of  
12 underwater vehicles such as torpedoes.

13 In light of the above, it is therefore understood that  
14 \_\_\_\_\_, the invention may be  
15 practiced otherwise than as specifically described.

1 Navy Case No. 77327

2  
3 FUEL DELIVERY SYSTEM

4  
5 ABSTRACT OF THE DISCLOSURE

6 A engine fuel delivery system for vehicles under test, which  
7 effectively limits the quantity of fuel under pressurization at  
8 any given time during a test sequence of the vehicle's engine.  
9 The system thus effectively reduces the magnitude of an explosion  
10 which may be caused by pressurization of fuel, and thus the scale  
11 of any damage, associated with any inadvertent detonation of  
12 pressurized engine fuel.

