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AFAL-TR-75-231



# A THOME AND ALLOS LABORATION

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# ADVANCED SLIT DETECTORS

HONEYWELL RADIATION CENTER HONEYWELL CORPORATE RESEARCH CENTER



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DECEMBER 1976

TECHNICAL REPORT AFAL-TR-75-231 FINAL REPORT FOR PERIOD 15 APRIL 1974 through 12 SEPTEMBER 1975

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exposure before delivery. Each detector was characterized for sensitivity, noise, dark resistance, speed of response and spectral response prior to and after each environmental exposure. Overall, these detectors have clearly demonstrated the capability of high sensitivity, sufficient speed and high tolerance to severe environmental conditions for future use in a strapdown stellar sensor.

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FOREWORD

This is the Final Technical Report of the GaP Environmental Testing Program. This program was sponsored by the Air Force Systems Command, Air Force Avionics Laboratory, Wright-Patterson Air Force Base, Ohio, under Contract No. F33615-74-C-1121. This work was performed during the period April 15, 1974 through September 12, 1975. The Project Engineer was Dr. R. A. Rotolante. Principal investigator was Dr. A. Chiang. Research Scientists were Drs. F. Pribble, P. Petersen and R. Schulze. The Program Manager was J. R. Farrell, Acknowledgements are due to Air Force Cambridge Research Laboratory, where the gamma ray exposure, the electron, proton irradiation and vibration tests were performed. Thanks are also due to Mrs. M. Young who fabricated most of the slit detectors and Mr. R. Healey who obtained most of the test data. The Program Contract Monitors, Mr. Larry Reitz and Mr. Charles Ennis, provided considerable technical direction and supporting data during the course of the program.

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## SECTION I

#### INTRODUCTION

This is the final technical report under AFAL Contract No. F33615-74-C-1121. The specific objectives of this program were to fabricate, characterize and deliver forty gallium phosphide photoconductive slit detectors  $(0.05 \times 0.0005 \text{ inch}^2)$  and subject 35 of these detectors to a series of environmental tests; temperature cycle, humidity, vibration, solar exposure, thermal vacuum, electron irradiation, proton irradiation, and gamma ray exposure before delivery. Each detector was characterized for sensitivity, noise, dark resistance, speed of response and spectral response prior to and after each environmental exposure.

The work effort under this contract resulted in the production, test and delivery of forty high performance photoconductive gallium phosphide detectors. The high sensitivity detectors were fabricated from Cu-doped, n-type GaP single crystals grown from solution. Copper is a deep lying acceptor in GaP which can be used to produce compensated material of high resistivity and acts as a sensitizing center. These centers are minority carrier traps which result in a longer electron lifetime and high photoconductive gain. The details of operation and initial development of Cu-doped GaP photoconductors are described in Reference 1. The Cu-doped GaP slit detectors were shown to be capable of measuring a laboratory simulated AO star of +7.5 magnitude (2 1/2-inch clear aperture) with signal to rms noise ratio of 17 which corresponds to an NEP of  $5 \times 10^{-16}$  watts//Hz. Detector rise and fall times less than 100 ms have also been observed. Most of the 35 detectors subjected to the environmental tests were able to operate without any degradation after severe environmental tests. The only test failures attributed to a detector occurred at the last stage (290°F) of the temperature cycle and two other failures occurred at the humidity test. Both of them resulted from the low melting point of indium which was used as an interconnection material between the contact metal (Te/Ag/Ni) and the gold wire bonds. Overall, these detectors have clearly demonstrated the capability of high sensitively, sufficient speed and high tolerance to environmental conditions for future use in a strap-down stellar sensor.

The material requirements and preparation and device fabrication and characterization are discussed in detail in Section II of this report. The results of the environmental tests are presented in Section III.

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Some of the device development and material growth effort reported in this document was under AFAL Contract No. F33615-C-75-1041.

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#### SECTION II

#### TECHNICAL DISCUSSION

The technical discussion is divided into three major sections; material requirement of Cu-doped GaP for star sensor applications, detector fabrication and testing procedures, and finally detector characterization.

GaP is a large bandgap semiconductor material (Egindirect = 2.25 eV). Cu-sensitized GaP operating in the photoconductive mode yields a high sensitivity device because of the photoconductive gain due to the trapping of minority carriers. The first part of this section discusses the major material parameter considerations in the design of detectors which are capable of meeting the program goal. Under AFAL Contract Nos. F33615-74-C-1121 and F33615-75-C-1041, a photo-lithographic technique was developed for the fabrication of gallium phosphide photoconductive slit detectors (0.05 x 0.0005 inch<sup>2</sup>) from Cu-doped material. A detailed description of detector fabrication and the detector preselection procedures will be found in Section 2.2. Finally, methods used to characterize the performance of detectors are described in Section 2.3.

2.1 MATERIAL REQUIREMENT AND PREPARATION

2.1.1 Material Requirement

During the program, Cu-doped solution grown GaP material was used to fabricate highly sensitive photoconductive slit detectors. Copper is a deep lying acceptor which can be used to produce compensated material of high resistivity and acts as a sensitizing center. (1,2,3,4) The sensitizing centers are minority carrier traps which result in a long electron lifetime and high photoconductive gain.

The performance design goal of the slit detectors for the star sensor application under AFAL Contract No. F33615-75-C-1041 was a peak signal to rms noise ratio of 10 to 1 for a signal energy of 6 x  $10^{-14}$  watts (integrated detector energy). A tradeoff between the material resistivity and the detector photoconductive gain requirements such that the detector is capable of meeting the design goals, has been carried out as follows. The signal current can be written as:

$$i_{g} = \eta F q G$$

where

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- $\eta$  = quantum efficiency of the device
- F = number of incident photons per second (1.5 x  $10^5$  ph/s at 0.5  $\mu$ m based on the spec)
- G = photoconductive gain

Let us consider the case that the detector thermal noise is the dominant noise source, the noise current per unit bandwidth can be written as:

$$i_{\rm N} = \left(\frac{4kT}{R_{\rm D}}\right)^{1/2}$$

where  $R_D$  is the detector resistance. It is straightforward to calculate that for  $R_D = 10^{10}$  ohm and  $\eta = 0.5$ , a gain of one will meet the performance specification. However, if  $R_D = 10^7$  ohm, a gain of 33 is required. The photoconductive gain required to meet the specification vs detector resistance is plotted in Figure 1, assuming a detector thermal noise limited case. It is easy to see the importance of using high resistivity\* material which will yield a detector with lower detector noise and therefore less gain required to meet the design goal.

#### 2.1.2 Material Preparation

Under AFAL Contract Nos. F33615-74-C-1121 and F33615-74-C-1041, ten copper-doped gallium phosphide (GaP) growth runs, AD-18 through AD-27 were completed and evaluated via Hall and photoelectric measurements.

The relation between the detector dark resistance and resistivity of the bulk material is based on the relation  $R_D = 2.8 \rho_D$ ; see the derivation in Appendix A.



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Figure 1 TRADEOFF BETWEEN DETECTOR PHOTOCONDUCTIVE GAIN AND DETECTOR RESISTANCE

Recrystallization AD-18 produced the first high resistivity n-type material resulting from recrystallizations performed under the current program. The run utilized starting GaP from Imanco Czochralski ingot one with a copper addition of 5.0 mole percent. Electrical measurements indicated the material was n-type with resistivities in the order of  $1 \times 10^6$  to  $1 \times 10^8$  Q-cm. Subsequent photoelectric measurements upon detectors fabricated from the material possessed gains of 80-90 with a response time less than 1 second.

Recrystallization AD-19 utilized starting material (Monsanto, lot one, 0.7 mole % Cu) which had been used previously and yielded p-type 100  $\Omega$ -cm material. Hall data obtained from AD-19 allowed the calculation of the activation energy of the material. The value obtained indicated a departure from the values of the activation energies obtained for earlier recrystallized materials.(1) The low value of the activation energy indicates that an acceptor in addition to the copper was introduced into the crystal structure.

Run AD-20 was intended to be a duplication of recrystallization AD-18 which yielded n-type material possessing a resistivity of  $1 \times 10^6 - 1 \times 10^8$   $\Omega$ -cm. Sections of Imanco ingot one were used as starting material with a copper addition of 5.0 mole percent. The resulting GaP was p-type with a resistivity of approximately 1000  $\Omega$ -cm. Initially, it was concluded the transition from n-type to p-type was rather abrupt. However, additional Hall measurements and calculation of the activation energy of the material again indicated an abnormal value.

Recrystallization AD-21 was the first copper-doped run employing lot two Monsanto starting material. A previous undoped recrystallization (AU-16) indicated a residual carrier (donor) concentration of ~5 x  $10^{15}$  cm<sup>-3</sup>. Run AD-21 utilized a copper addition of 0.4 mole percent. The resulting material was p-type, possessing a resistivity of ~100  $\Omega$ -cm. Again the activation energy deviated from those possessed by materials from recrystallizations prior to run AD-19. The additional acceptor has not been identified but could possibly be zinc. The procedures were analyzed, but the origin of the possible zinc contamination was not apparent.

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In recrystallization AD-22, an attempt was made to approach compensation of the Imanco ingot material from the n-type side. The copper addition was lowered from 5.0 mole percent to 4.0 mole percent. The resulting material was n-type possessing resistivities over the range of 1 x  $10^3$  to 1 x  $10^{10}$  Ω-cm. The crystals produced were small in size.

The yield of larger size crystals appears to be influenced by the rate of growth and temperature range over which precipitation occurs. Slower precipitation rates and longer growth periods (wider temperature range of precipitation) are more efficient in producing large crystals. The recrystallization AD-23 utilized the second lot of Monsanto polycrystalline GaP as starting material. The residual carrier concentration was inferred from Hall measurements performed on a crystal from an undoped run (AU-16). The data indicated a carrier (donor) concentration of approximately 5 x  $10^{15}$  cm<sup>-3</sup>. The copper concentration employed in run AD-23 was 0.2 mole percent and an extended temperature range of precipitation was utilized (1150-1000°C). The resulting material was p-type possessing a resistivity of approximately 1 x  $10^3$   $\Omega$ -cm. The consistent growth of p-type material when employing Monsanto lot two starting GaP and activation energy calculations tend to indicate an additional unknown acceptor is being introduced during the growth process. Possibly, it is most evident in the Monsanto material because of its low residual carrier concentration. If the unknown acceptor was being inadvertently introduced in quantities equal to the concentration of the residual carrier, p-type material would be expected. The addition of copper will assure p-type material. Previous data (although minimal) tended to indicate the introduction of the unknown acceptor was associated with the addition of copper to the melt. Undoped recrystallizations employing Monsanto GaP yield n-type material with expected activation energies.

Recrystallizations AD-24 and AD-25 had similar growth parameters. Copper was added in the amount of 4.2 mole percent. The growths were terminated (air quenched) at 1080°C. The recrystallizations employed starting material from Imanco Czochralski ingot one. Preliminary data indicated material of high resistivities ( $10^6 - 10^8 \Omega$ -cm). Photoelectric measurements on a few samples indicated satisfactory photodetector characteristics.

Recrystallization AD-26 is the initial recrystallization employing a new starting material (Imanco Czochralski ingot two). The Hall characteristics of the two ingots are not identical. Ingot two possessed a somewhat higher residual carrier concentration and a lower resistivity than ingot one. Further, optical

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absorption data indicated nearly equivalent sulfur and silicon concentrations. Ingot one possessed sulfur as the primary impurity (material was nominally undoped) with silicon being present in concentrations about an order of magnitude lower  $(2-4 \times 10^{15} \text{ cm}^{-3})$ .

The quantity of copper added to the solution in recrystallization AD-26 was 4.2 mole percent. The starting material was taken from the seed end of the ingot to obtain a residual carrier concentration (~5.0 x  $10^{16}$  cm<sup>-3</sup>) near that (3.9 x  $10^{16}$  cm<sup>-3</sup>) of the seed end of ingot one. Seed end material from ingot one yielded two consecutive runs of photodetector quality material (AD-24, AD-25).

Recrystallization AD-26 yielded n-type material of which two Hall samples indicated resistivities of 10  $\Omega$ -cm and 50  $\Omega$ -cm (starting material resistivity = 0.7402  $\Omega$ -cm). Since the resistivity of the material was low, it was concluded the material did not have adequate photoresponse for the fabrication of detectors.

The AD-27 run was accomplished to bracket the concentrations of copper required to produce n- and p-type material. Recrystallization AD-26 yielded low (10-50  $\Omega$ -cm) resistivity n-type material while AD-27 yielded high (1 x 10<sup>4</sup>  $\Omega$ -cm) p-type copper-doped gallium phosphide. These data suggest that 7.5 mole percent copper is near the concentration required for close compensation of ingot two material.

Recrystallization AD-27 differed from recrystallization AD-26 in that temperature instabilities were introduced during the first 48 hours of growth and a temperature rise (about 2°C) was inadvertently introduced during the early part of the growth cycle (shortly after the 48-hour period). The predominant type of growth in AD-26 was surface growth while in AD-27, the growth was confined to one end of the growth container and the crystals formed in the interior of the solution. Run AD-27 yielded two larger crystals (irregular shaped, approximately 0.250" x 0.375")

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and a number of somewhat smaller crystals, even though the growth was quenched at a relatively high temperature (1078°C).

Hall samples (two) were prepared from the large crystals and measurements indicated similar resistivities (~1 x  $10^4 \ \Omega$ -cm).

2.2 DETECTOR FABRICATION AND TESTING PROCEDURES

2.2.1 Detector Fabrication

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During these two programs, a photolithographic procedure was developed to fabricate a radial GaP array (detector element size:  $0.05 \times 0.0005$  inch<sup>2</sup>). The fabrication procedures used in the present program were evolved from those used in Phase I and Phase II programs. Several improvements were developed. A sequence of the fabrication procedure is shown schematically in Figure 2.

The as-received Cu-doped gallium phosphide crystal material is in the form of platelets and prism-shaped pieces. In step 2, Figure 2, the thickness uniformity over the surface area (back and front) of the pieces is achieved by mechanical lapping procedures. The crystal pieces are supported with beeswax on stainless steel lapping blocks during the lapping operation. Sequential lapping of the surface was performed with slurries of W-8 garnet. The final lapping of the surface was carried out on a suspension of  $3-\mu m$  aluminum oxide particles in water and glycerine. Relative smooth and pitless pieces with parallel surfaces were obtained in this manner. A sequence of organic solvents and detergent solutions were used to remove the residual wax film and adherent abrasive particles on the surface. The surface damage left after the mechanical lapping operation is removed by chemical etching, shown in step 3. The etch solution used is a diluted aqua regia (2 HC1:2H<sub>2</sub>0:1 NHO<sub>3</sub>). The etch was terminated by quenching with deionized water.

The ohmic contact formation is conducted in steps 4 and 5 of Figure 2. In step 4, a closely spaced (0.0005 inch) evaporation mask which defines the detector active area is placed in contact with the p-face of the crystal material resting in a holder. The entire fixture is then placed in a vacuum chamber



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and a high vacuum (<  $2 \times 10^{-6}$  torr) established. The electrode alloy is then deposited by evaporation from a source held about 6 inches from the evaporation fixture. Two evaporation sources are used for each evaporation. The first is a molybdenum boat containing Ag-1% Te alloy source. The second is a tungsten boat containing nickel wire. Approximately 4 Å to 6Å of tellurium is first evaporated and then the temperature of the molybdenum boat is increased to permit the evaporation of silver, about 600 Å. At this point, power to the boat is turned off. Nickel is then evaporated from the second evaporation. About 600 Å of nickel are evaporated.

Sintering of the evaporated alloy, step 5, was also carried out in a vacuum. The detectors are located next to a thermalcouple placed in an aluminum oxide coated molybdenum boat. The condition adopted for successive sintering was to increase the boat temperature (thermal couple reading) to 520°C in 30 minutes, and to maintain it at 520°C for 20 minutes. After sintering, any excess metallic contaminant ions in the interelectrode area are removed. In step 7, the active area is then covered by a protective coating by using a light field photomask and Shipley photoresist. A layer of indium  $(1/2 \text{ to } 1 \mu \text{m})$  is evaporated over the contact regions as shown in step 8 of Figure 2. In steps 9 and 10, the photoresist is removed and a 1-mil gold wire is thermal compression bonded to the In bonding pad. Finally, the device is attached with temporary adhesive to a flat pack and the gold wires are bonded to the terminals of the board. The device is ready for test.

#### 2.2.2 Detector Preselection and Testing Procedures

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During the star sensor and environmental testing programs, we followed the testing schedule shown in Figure 3. After Ag-1% Te and Ni contact evaporation and sintering, Only if the sample shows light the detector is probed. sensitivity under a microscope lamp, is the indium bonding pad then evaporated and the finished device mounted on the package for further testing. The dark current-voltage measurement is the first testing procedure to check if an ohmic contact has been achieved. The device is then tested for its spot scan response using a GFE star transit simulator. If the measured sensitivity of the device falls within a specified range of the design goal, then the detector optimum bias point, rise and fall time, and wideband noise are measured. Spectral responses of selected devices of each fabrication run are taken. After the device is fully characterized, it is ready for environmental test. Thirty-five



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selected slit detectors separately underwent the following tests: temperature cycle (4 detectors), humidity (5 detectors), vibration (4 detectors), solar exposure (5 detectors), thermal vacuum (8 detectors), electron irradiation (3 detectors), proton irradiation (3 detectors), gamma rays (3 detectors). Detectors were checked for sensitivity, noise, and rise and fall times after each environmental exposure. Any change or failure which occurred during the environmental testing was reported. The cause of failure was studied and corrective action was initiated to prevent reoccurrence of such a failure. The 35 detectors were then delivered to AFAL.

#### 2.3 DETECTOR CHARACTERIZATION

The following methods were used to characterize the performance of the Cu-doped GaP slit detectors:

- Dark I-V measurement
- Spot scan sensitivity
- Noise measurement
- Rise and fall time
- Spectral response

#### 2.3.1 Dark Current-Voltage Measurement

The current-voltage relation of each detector was measured with a Keithley electrometer. A battery bias voltage source, with a potentiometer, was used to set the bias voltage. Most detectors exhibited close to a perfect ohmic relationship. Typical examples are shown in Figures 4 and 5.

#### 2.3.2 Spot Scan Sensitivity

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The spot scan results were measured with a GFE star transit simulator. The source used was a 9-amp GE ribbon filament lamp operated from a controlled power supply. AO spectral filters (BG 34-3.5 mm, KG 2-3.0 mm) and a set of neutral density filters could be inserted into it to simulate the desired color temperature and appropriate stellar magnitude. A small circular aperture was inserted into the optic path before the image lens. The image was a 1.5-mil diameter dot. The focus was adjusted to achieve maximum signal. A calibrated IL selenium detector (0.25  $\mu$ m to 0.7  $\mu$ m), an IL radiometer and a calibrated EG&G Si detector (0.4-1.0  $\mu$ m) were used to calibrate the laboratory source. When





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there is 8-amp current on the 9-amp lamp with BG and KG filters inserted, and ND filter setting at 53, the spectral irradiance of the source is:

Wavelength (nm)Spectral Irradiance (W/nm)399.5 $2.5 \times 10^{-13}$ 489.2 $4.3 \times 10^{-13}$ 549 $2.9 \times 10^{-13}$ 599 $2.3 \times 10^{-13}$ 

The spectral irradiance of an AO zero magnitude star (2-1/2 clearaperture) and the laboratory simulated source are plotted in Figure 6. The total optical power of the simulated source between 400 and 500 nanometers is about  $1.5 \times 10^{-11}$  watts. Because the diameter of the image spot is 1.5 mil and the detector slit width is only 0.5 mil, the actual optic power which is incident on the detector active area is only 8.6  $\times 10^{-12}$  watts. Throughout this final report, this test condition will be referred to as AO zero magnitude star. With the aid of calibrated neutral density filters, different stellar magnitudes can be obtained.

After dark I-V measurement, each detector was tested for its spot scan response. Photographs of scope traces during spot scan across the slit from left to right and then reverse direction were taken, which resulted in two signal peaks (see Figure 7). The signal output was taken from a transimpedance amplifier. The schematic and the gain profile of the amplifier are shown in Figures 8 and 9 respectively. The calibrated AO zero magnitude source and 0.1% transmission neutral density filter were used to simulate a  $\div$ 7.5 m(V) star. Unless otherwise specified, the scanning rate used in this test is always 0.00075 inch/s.

#### 2.3.3 Noise Measurements

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Each detector was tested for its wideband noise. The sequence is as follows. Connect detector noise output through the transimpedance amplifier (shown in Figure 8). Display the output on a storage oscilloscope with the time base set for 1 to 2 ms/cm with the trace free running. After about 5 seconds,





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-	AD25 M7 NO. 1
=	+7.5 m(V) AO
=	28 m V
=	0.00075 in./s
-	2.35 x $10^{10}$ ohm
-	-15 V
	-

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Figure 7 SPOT SCAN SENSITIVITY



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Figure 8 GaP PREAMPLIFIER



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photograph the noise envelope with appropriate shutter speed. The photo will show the noise envelope during the storage period. Peak-to-peak noise voltage,  $V_{N, pk-pk}$  is taken to be the voltage difference between the extreme positive and negative deflection of the noise envelope. A sample of wideband noise of the slit detectors is shown in Figure 10.

The noise spectra of selected detectors were taken through the same amplifier circuit. We found that 1/f noise is a dominant noise source in most devices (see Figures 11 and 12). Surface recombination may be the origin of the excess 1/f noise in those slit detectors. The existence of depletion and inversion layers caused by interface states is the other possible explanation for the noise. Further investigation is required to understand and then to eliminate the excess noise.

#### 2.3.4 Rise and Fall Time

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A mechanical shutter was placed in the optical path of the spot scan setup. The spot was positioned on the detector active area. The rise time of the detector was obtained by opening the shutter and then measuring the time from the 10% to 90%point of the saturated dc shift of the output signal when the detector was dark and under illumination. Then the shutter was closed and again the fall time from the 10% to 90% point was taken. We observe, in general, there are two distinct rise times of the Cu-doped slit detector (see Figure 13): one very fast signal rise to the flat signal level plus a very slow rise to the saturated peak signal level. The observed slow response is caused by the gradual filling of the majority  $traps^{(5)}$ . On the same figure the device has two distinct fall times; an initial fast fall time due to the recombination mechanism, followed by a long decay resulting from thermal emptying of the traps.

The photoresponse of the GaP:Cu sample to pulsed radiation was measured by Honeywell and AFAL. In Figure 14, the signal level varies as a 100-ms light pulse is repeated on the detector in 10 second intervals. Prior to the first pulse the detector had been stored in total darkness for 85 hours. At room temperature, all the majority traps are empty. During the first pulse, part of the photon generated electrons from the valence band are captured by the electron traps which yield a smaller signal pulse. The effect of majority traps is apparent, as evidenced by the increase of the height of the second pulse over the first.



Detector ID	2.5	AD18 M7 NO. 4
V <sub>Np·p</sub>	<u></u>	4 mV
R <sub>F</sub>	1	2.3 x $10^{10}$ ohm
V <sub>B</sub>	-	-10 V

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### Figure 10 DETECTOR WIDEBAND NOISE



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Rise Time



Fall Time

Detector ID = AD18 M7 NO. 8  $V_{BIAS}$  = -10 V  $R_F$  = 2.35 x 10<sup>8</sup> ohm Source = +2.5 m(V) (AO + 10% ND Filter) Figure 13 DETECTOR RISE AND FALL TIME



1st Pulse





(a)



30 s after Start of Chopped Radiation 8 min after Start of Chopped Radiation (Saturation)

(b)

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Figure 14 (a) TIME-DEPENDENT RESPONSE OF A GaP:Cu SAMPLE TO A SERIES OF 100-ms PULSES AT 10-s INTERVALS OF 2.8-eV RADIATION. (b) RESPONSE OF THE SAME SAMPLE TO A REGULARLY MODULATED SOURCE.
Pulse response data of the Cu-doped GaP slit detector were taken at AFAL and shown in Figure 15. As compared to Figure 14, the height of the pulse response is slowly decreasing to a steady value. This may be due to the presence of deep lying acceptor levels, which have a large cross section with respect to holes. The detector had been stored in the dark for 12 hours; all the levels are ionized (contain no holes) in the dark. The free holes created during the first pulse are captured by the empty acceptor levels which lead to a longer electron lifetime and higher signals. After the first pulsed excitation, because of the high binding energy of those levels, part of the captured holes will stay in these levels. During the second pulse, because part of the hole traps are already occupied, more free holes are produced in the valence band. The net effect of those free holes is an increase in electron recombination and yields a decrease in pulse height.

### 2.3.5 Spectral Response

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The spectral response of the selected slit detectors was measured by using a tungsten source and a Jarrel Ash 0.25 meter Ebert monochrometer. The peak response of those detectors is around 0.5  $\mu$ m, as shown in Figures 16, 17 and 18. Because of the high absorption coefficient ( $\alpha > 10^4$ cm<sup>-1</sup>)<sup>(6)</sup> of GaP at photon energies greater than 3 eV, the fast decrease of the spectral response at shorter wavelengths ( $\lambda > 0.4 \mu$ m) is indicative of high surface recombination. An improvement in surface preparation during the fabrication process in needed to give better response of high energy photons.

### 2.3.6 Photoconductive Gain Measurement

The photoconductive gain is defined as the ratio of the number of charge carriers passing through the external circuit per second to the number of photons incident on the detectors per second. The photoconductive gain was obtained by measuring the dc shift of the output signal both when the slit detector is dark and under the illumination of a 1.5-mil dot. A sample calculation is shown as follows:

> Detector: AD18M1A No. 2 Incident flux: 1.7 x  $10^{-7}$  watts/cm<sup>2</sup> at 0.489  $\mu$ m Bias voltage: 20 V  $R_F = 2.3 \times 10^{10}$  ohms Signal through amplifier = 250 mV A = slit area under illumination = 4.85 x  $10^{-6}$  cm<sup>2</sup>



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Figure 15 DETECTOR PULSE RESPONSE







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Gain = 
$$\frac{260 \times 10^{-3}}{2.46 \times 10^{18} \times 1.7 \times 10^{-7} \times 4.86 \times 10^{-6} \times 2.3 \times 10^{10} \times 1.6 \times 10^{-19}} = 35$$

If the illumination is positioned in a different active area of the same detector, photoconductive gain as high as 300 has been observed. See Table 1.

2.3.7 Detector Uniformity

Spot scan response along the detector active area was measured. It indicates that lack of spot scan uniformity along the slit is the chief drawback of those devices (see Figures 19 and 20). As stated in the beginning of this report, for the photoconductive mode detector, very high purity, high resistivity, n-type material is required. Considerable material growth development effort is still required to achieve accurate compensation and uniformity.

### 2.3.8 Linearity of the Detector Response

Measurements of the dependence of spot scan response on the excitation energy were made. One can see in Figures 21 and 22, the response vs light intensity is linear in the energy range under investigation. Similar tests were taken by AFAL on five delivered slit detectors (see Figure 23). Over four orders of magnitude linear dynamic range was obtained.

### 2.3.9 Optimum Bias Point

The spot scan response and pk-pk noise data of each detector was measured under different bias conditions. An optimum bias point corresponding to the maximum signal and noise ratio was selected for each detector. A bias dependent signal and noise measurement is plotted in Figure 24.

### 2.4 DETECTOR PERFORMANCE SUMMARY SHEET

The performance of the 35 slit detectors before the environment tests are summarized in Table 2.

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	Spot Scan Sensitivity mV	Rise Time	Saturated dc Response mV	Photoconductive Gain (nG)	Majority Carrier Lifetime* ns
20		100 ms	20	2.6	5.2
38		1.2 s	45	5.8	12
30		700 ms	30	3.9	œ
32		7 s	175	22.6	48
50		1.5 s	80	10.4	20
20		100 ms	20	2.6	5.2
20		100 ms	20	2.6	5.2
1000		85	2500	323	650
Majorit) assuming	r carrier g ημ = 40	lifetime is	calculated fro	m the saturated phot	ocurrent



Figure 19 SPOT SCAN UNIFORMITY ALONG THE SLIT LENGTH

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### DETECTOR RESPONSE VS POSITION





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Figure 21 DETECTOR RESPONSE VS INPUT POWER



Figure 22 DETECTOR RESPONSE VS INPUT POWER

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AFAL DATA



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Figure 23 GaP LINEARITY DATA

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 Table 2

 35 GaP SLIT DETECTORS PERFORMANCE SUMMARY CHART

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	Environmental	Dark Resistance (ohm)	Spot Scan Sensitivity	Noise Vn pk-pk	Optimum Bias	Rise Time	Fall Time	R <sub>F</sub> (ohm)
PERCENT IN	acter 13	2 5 × 10 <sup>8</sup>	50 mV at 0 m(V)	600 LA	-15 V	200 ms	100 ms	3 × 10 <sup>7</sup>
ADIG-LA NO. 10	Taperature	1 1 + 109	70 mV at +2.5 m(V)	4 mV	-5 V	•		ł
ADIG-16 No. 1	Temperature	7.7 × 10 <sup>7</sup>	14 mV at +2.5 m(V)	1 mV	ı	1		•
AD18-L9 No. B	Gamma	1.4 × 10 <sup>8</sup>	40 mV at +2 m(V)	3.8 mV	+13.5 V	60 ms	1	3 × 10'
AD18-L11 No. 2	Temperature	$6 \times 10^7$	5.6 mV at 2.5 m(V)		•	•	•	•
AD18-L12 No. 1	Electron	1.8 x 10 <sup>9</sup>	30 mV at 0 m(V)	500 uV	-5V	200 ms	50 ms	1 × 10 <sup>6</sup>
AD18-L16 No. 1	Humidity	4.6 x 10 <sup>8</sup>	120 mV at 0 m(V)	20 mV	+10 V	10 ms	80 ms	3 × 10 <sup>7</sup>
AD18-L16 No. 2	Humidity	4.7 x 10 <sup>8</sup>	60 mV at +0.75 m(V)	20 mV	+10 V	80 ms	80 ms	3 × 10 <sup>7</sup>
AD18-L17 No. 1	Gamma	$4.2 \times 10^{10}$	25 mV at +2 m(V)	2 mV	-5 V	250 ms	400 ms	$1 \times 10^{10}$
AD18-L17 No. 2	Gamma	$1.4 \times 10^{10}$	120 mV at +2 m(V)	4 mV	+15 V	80 ms	80 ms	4 x 10 <sup>9</sup>
AD18-M7 No. 1	Thermal Vac	$6.4 \times 10^{11}$	32 mV at +2.5 m(V)	4 mV	v 7-	170 ms	200 ms	2.3 × 10 <sup>10</sup>
AD18-M7 No. 2	Thermal Vac	$3.0 \times 10^{11}$	120 mV at 0 m(V)	8 mV	v 7-	100 ms	<b>sm</b> 06	2.3 × 10 <sup>10</sup>
AD18-M7 No. 3	Thermal Vac	$3.6 \times 10^{11}$	160 mV at +1.2 m(V)	6 mV	-10 V	120 ms	100 ms	2.3 × 10 <sup>10</sup>
AD18-M7 No. 4	Thermal Vac	$2.5 \times 10^{11}$	38 mV at +2.5 m(V)	4 mV	-10 V	300 ms	300 ms	2.3 × 10 <sup>10</sup>
AD18-M7 No. 5	Proton	$1.2 \times 10^{11}$	50 mV at +5 m(V)	5 mV	+5 V	50 ms	70 ms	2.3 × 10 <sup>10</sup>
AD18-M7 No. 6	Proton	$2.3 \times 10^{11}$	80 mV at +1.2 m(V)	6 шV	+5 V	90 ms	150 ms	2.3 × 10 <sup>10</sup>
AD18-M7 No. 7	Proton	$5.4 \times 10^{11}$	460 mV at +5 m(V)	6 mV	-7 V	2.2 s	120 ms	2.3 × 10 <sup>10</sup>
AD18-M7 No. 8	Thermal Vac	2.8 × 10 <sup>9</sup>	50 mV at +2.5 m(V)	6 mV	-10 V	800 ms	100 ms	2.3 × 10°
AD18-M7 No. 9	Thermal Vac	$7 \times 10^{11}$	600 mV at +5 m(V)	8 mV	+10 V	800 ms	200 ms	2.3 × 10 <sup>44</sup>
AD18-M8 No. 1	Solar	$1.3 \times 10^{10}$	360 mV at +5 m(V)	30 mV	-10 V	l s	300 ms	$1 \times 10^{10}$
AD18-M8 No. 2	Solar	8.3 x 10 <sup>10</sup>	120 mV at +2.5 m(V)	18 mV	-16 V	1.2 s	200 ms	4.3 × 10 <sup>7</sup>
AD18-M8 No. 3	Solar	4.2 × 10 <sup>9</sup>	80 mV at 0 m(V)	18 mV	+10 V	500 ms	300 ms	9 x 10 <sup>8</sup>
AD18-M9 No. 4	Vibration	4.1 x 10 <sup>8</sup>	140 mV at +2.5 m(V)	5 mV	+15 V	40 ms	40 ms	$3 \times 10^7$
AD18-M9 No. 6	Vibration	2.7 × 10 <sup>9</sup>	100 mV at m(V)	5 mV	+15 V	400 ms	100 ms	3 × 10'
Cu9-16	Temperature	$1.4 \times 10^7$	10.5 mV at +2.5 m(V)	2.2 mV	•	•	1	,
AD24-L1 No. 1	Thermal Vac	1.9 × 10 <sup>8</sup>			ı	,		•
AD24-L1 No. 4	Thermal Vac	5.3 × 10 <sup>4</sup>	ı	,	ł	ı	•	۵ ۱
AD25-M3 No. 1	Electron	$4.0 \times 10^{9}$	2.5 V at +2.5 m(V)	3.2 mV	-6 V	100 ms	100 ms	2.3 × 10°
AD25-M5 No. 1	Vibration	2.5 × 10 <sup>7</sup>	300 mV at +2.5 m(V)	S mV	-10 V	800 ms	500 ms	
AD25-M5 No. 2	Vibration	$2.0 \times 10^{8}$	45 mV at +2.5 m(V)	2 mV	-5 V	800 ms	1.4 s	•
AD25-M6A No. 1	Humidity	$1.0 \times 10^{10}$	700 mV at 1.2 m(V)	16 mV	-15 V	1.1 s	8	9 x 10°
AD25-M6A No. 2	Humidity	3.9 × 10 <sup>10</sup>	400 mV at 0 m(V)	12 mV	+15 V	400 ms	l s	9 × 10 <sup>0</sup>
AD25-MGA No. 5	Humidity	$4.0 \times 10^{10}$	200 mV at +5 m(V)	S mV	+5 V	250 ms	250 ms	1 × 10 <sup></sup>
AD25-M7 No. 1	Solar	2.0 × 10 <sup>11</sup>	30 mV at +7.5 m(V)	S mV	-15 V	600 ms	260	2.3 × 10
AD25-M7 No. 2	Solar	4.4 × 10 <sup>8</sup>	30 mV at +2.5 m(V)	6 mV	+15 V	400 ms	240 ms	3 × 10'

### SECTION III

### ENVIRONMENTAL TEST RESULTS

Under AFAL Contract No. F33615-74-C-1121, 35 gallium phosphide photoconductive slit detectors fabricated from Cu-doped material were subjected to a series of environmental tests; temperature cycle, humidity, vibration, solar exposure, electron irradiation, proton irradiation and gamma ray exposure. Detectors were checked for sensitivity, noise, and rise and fall times before and after each environmental exposure. Any changes in detector performance were noted. The detailed test conditions are described below.

### 3.1 TEMPERATURE CYCLE

Four detectors were exposed to  $0^{\circ}$ F,  $100^{\circ}$ F,  $200^{\circ}$ F and  $290^{\circ}$ F for 10 hours each. One detector was capable of operating without degradation after each increment of temperature. The other three detectors passed at 200°F exposure. The indium bonding pads of the devices started to melt and shorted the active area at 290°F. GaP is a large bandgap semiconductor and is capable of operating at high temperature. GaP rectifier operating at 500°F has been reported <sup>(6)</sup>. For the purpose of higher device storage and operating temperature, a high melting point metal such as gold is recommended for use as a bonding pad.

The device performance before and after each temperature exposure is summarized in Table 3.

### 3.2 VIBRATION

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Four detectors (AD18 M9 No. 4, AD18 M9 No. 6, AD25 M5 No. 1, and AD25 M5 No. 2) were capable of operating without degradation after exposure to the following vibration levels on each of the three mutually perpendicular axis

Frequency Range (Hz)	G-Level
10 - 14	2.5g
14 - 40	5.0g
40 - 400	15.0g
400 - 3000	25.0g

This test was performed at Air Force Cambridge Research Laboratories.

Table 3

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### TEMPERATURE CYCLE TEST RESULTS

ctor	Spot Sca Before	n Sensitivity After 200°F	[+1.2 m(V)] After 290°F	Before	After 0°F	After 200°F	After 290°F	kefore	Dark Ro Afrey 0°F	after 200°F	After 290°F
8-15	70 mV	70 mV	(1)	4 mV	4 mV	Ve 4	(1)	1.1×10 <sup>9</sup>	1.2×10 <sup>9</sup>	1.1x10 <sup>9</sup>	(1)
8-L6 0. 1	14 mV	14 mV	14 mV	1.2 mV	1.2 mV	1.2 mV	2.4 mV	7.7×10 <sup>7</sup>	9×10 <sup>7</sup>	7.7×10 <sup>7</sup>	7.7×10 <sup>7</sup>
8-L11 0. 2	5.6 mV	5.6 mV	(2)	0.5 mV	0.6 mV	0.7 mV	(2)	6×107	7×10 <sup>7</sup>	6x10 <sup>7</sup>	(2)
-L6	10.5 mV	10.5 mV	(1)	2.2 mV	2.2 mV	2.5 mV	(1)	1.4×107	1.5×107	1.4×10 <sup>7</sup>	(1)

(1) Indium started to melt. NOTES:

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### 3.3 HUMIDITY

The humidity tests were conducted at atmospheric pressure in a Hieatt chamber. Five bare detectors (two mounted on T05 cans, three mounted on flat packs made by Metallized Ceramic Corp) were stored in the chamber for 16 hours at  $102^{\circ}F \pm 2^{\circ}F$  and relative humidity at 90%  $\pm 4\%$  maintained by distilled water. After the exposure, two detectors with dark resistance ~ 10°  $\Omega$ , which were mounted on T05 can, were able to operate without degradation. The open terminal resistance of the flat pack on which the other three detectors were mounted was degraded to  $10^{\circ}$   $\Omega$ . After remount of the three detectors on a new flat pack, one detector was capable of operating without degradation. The other two detectors did not survive the remount process. The detector performance before and after the humidity test is summarized in Table 4.

### 3.4 SOLAR EXPOSURE

Five bare detectors mounted on flat packs were tested in direct sunlight. These measurements were made from 11:30 am to 2:30 pm on June 20, 1975 in Lexington, MA. The weather was clear with very light haze. The calculated incident solar energy from the sun angle at 12:30 pm for a perfectly clear day was 130 mW/cm<sup>2</sup> (based on AMO = 139 mW/cm<sup>2</sup>)<sup>(7)</sup>. Each detector was exposed to the solar radiation for periods of 1, 5, 10, and 15 minutes with recovery time noted after each increment time of exposure (see Figure 25). One can see that changing the duration of the solar exposure has little effect on the decay process. The decay process after solar exposure was characterized by an initial fast time constant and followed by a long decay time of several seconds.

There was no measureable change in detector performance before and after the solar exposure (see Table 5).

### 3.5 ELECTRON IRRADIATION

Three GaP:Cu slit detectors were evaluated before and after electron radiation. 1 MeV electron beam from a Van de Graaff accelerator at AFCRL was used. The beam intensity was set at  $0.265 \ \mu A/cm^2$ -s (1.66 x 10<sup>12</sup> e/cm<sup>2</sup>-s). After a 606-s exposure, a total fluence of 10<sup>15</sup> e/cm<sup>2</sup> was accumulated. For a 1 MeV electron beam, a dose rate of 1 e/cm<sup>2</sup>-s corresponds to 3.72 x 10<sup>-8</sup> rads/s. Therefore, a total dose of 3.72 x 10<sup>7</sup> rads (Si) was accumulated during the electron radiation test. Table 4

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### HUMIDITY TEST

	Dark Resista	nce D	Spot Scan F	Response	Noise V <sub>N</sub> , P	k-pk	Rise Tin	Ę	Fall Tim	a
Detector	Before	After	Before	After	Before	After	Before	After	Be fore	After
AD18-L16 No. 1	4.6 × 10 <sup>8</sup>	4.9 x 10 <sup>8</sup>	120 mV, at 0 m(V)	110 mV	20 mV	l5 mV	70 ms	70 ms	80 ms	100 ms
AD18-L16 No. 2	4.7 × 10 <sup>8</sup>	3.8 × 10 <sup>8</sup>	60 mV at +0.75 m(V)	60 mV	20 mV	14 mV	80 ms	80 ms	80 ms	80 ms
AD25-MGA No. 1	1 × 10 <sup>10</sup>	5 x 10 <sup>9</sup>	700 mV at +1.2 m(V)		16 mV	·	1.1 s		s C	
AD15-MGA No. 2	3.9 × 10 <sup>10</sup>	1.5 × 10 <sup>10</sup>	400 mV at 0 m(V)	i.	12 mV		400 ms		1 s	Π.
AD25-M6A No. 1	4.0 × 10 <sup>10</sup>	4.2 × 10 <sup>10</sup>	200 mV at +5 m(V)	180 mV	S av	5 mV	250 ms	250 ms	250 ms	250 ms

- Flat Packs Failed



Detector AD18M8 No. 3

Decay time after 1 min solar exposure



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Decay time after 5 min solar exposure

### Figure 25 DETECTOR DECAY TIME AFTER SOLAR EXPOSURE



Detector AD18M8 No. 3

Decay time after 10 min solar exposure



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Decay time after 15 min solar exposure

Figure 25 (continued)

### Table 5

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## DETECTOR PERFORMANCE BEFORE AND AFTER SOLAR EXPOSURE

	Dark Resist	tance A	Spot Scan	Response	Noise V <sub>N</sub> ,	pk-pk	Rise T	ime	Fall T	lave
Detector	Before	After	Before	After	Before	After	Before	After	Before	After
AD25-M7 No.1	2 × 10 <sup>11</sup>	2.4 × 10 <sup>11</sup>	30 mV +7.5	30 mV m(V)	5 mV	S BV	600 ms	700 ms	260 ms	240 ms
AD25-M7 No. 2	4.4 × 10 <sup>8</sup>	7.8 × 10 <sup>8</sup>	30 mV +2.5 m	30 mV (V)	6 BV	5 mV	400 ms	400 ms	240 ms	240 ms

The dark resistance, spot scan sensitivity and rise time of the three detectors before and after electron radiation are tabulated in Table 6. The most noticeable change is the detector rise time. Right after the electron exposure, the detector rise times are slower than before the electron radiation. Detector AD-25 M3 No. 1 was tested after being at room temperature for 70 hours. It exhibited less change than the two detectors which were tested immediately after the radiation. The dark resistance of those detectors measured within 70 hours after the electron radiation were about twice higher than that before. When measured four months after electron radiation, the dark resistance recovered to its initial value. We concluded that room temperature annealing caused this. In that case, long term dose accumulation an in-space application would result in even greater radiation tolerance.

### 3.6 PROTON IRRADIATION

Three detectors were used for proton irradiation test. The test was performed by using AFCRL's linear accelerator. The beam intensity was set on the order of  $10^{10}$  P/cm<sup>2</sup>-s and continuously monitored until a total dose of  $10^{11}$  P/cm<sup>2</sup> was accumulated. The beam uniformity and the alignment of the sample holder were checked before the run. The detectors were capable of operating after the irradiation without any degradation (see Table 7).

### 3.7 GAMMA RAY EXPOSURE

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Three detectors fabricated on this program were evaluated under gamma ray radiation.

To measure the gamma response of the GaP detectors, the detector and its preamplifier were placed inside a hot cell at AFCRL (Figure 26). A Victorian radiation-probe was positioned at approximately the same level and at approximately the same distance away from the source as the detector. This measured the gamma flux coming from the cobalt-60 source in terms of Roentgens per minute. The hot cell was sealed during the test and the detectors, mounted in T0-5 cans with open sleeve covers, were covered with black tape.

Preliminary checks were performed to assure that no noise was produced in the electronics or cabling while in a gamma environment.

Experiments were performed to characterize the response at room temperature of GaP photoconductors to 1 MeV gamma rays. Table 6

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# DETECTOR PERFORMANCE BEFORE AND AFTER ELECTRON FLUENCE OF $10^{15} \text{ e/cm}^2$

	Δ	ark Resistance G		Spot Scal	n Sensicivity	Noise (	pk-pk)	Rise T	ime
Detector	Before	Within 70 hrs. after irradiation	4 Month after irradiation	Before	Within 70 hrs. after irradiation	Before	Within 70 hrs. after irradiation	Before	After
ADI8-LA No. 1	2 × 10 <sup>8</sup>	5.6 x 10 <sup>8</sup>	2 x 10 <sup>8</sup>	50 mV 0	30 mV (V)	600 µV	600 µV	200 ms	ŝ
AD18-L12 No. 1	1.8 × 10 <sup>9</sup>	4.1 × 10 <sup>9</sup>	1.5 × 10 <sup>9</sup>	36 mV 0	22 mV m(V)	500 Juv	800 μν	200 ms	600 m
AD25-M3 No. 1	4 × 10 <sup>9</sup>	5 x 10 <sup>9</sup>	3.4 × 10 <sup>9</sup>	2.5 V 2.	1.8 V 5 m(V)	3.2 mV	4 mV	100 ms	200 ms

Table 7

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DETECTOR PERFORMANCE BEFORE AND AFTER PROTON FLUENCE OF  $10^{11}$  P/cm<sup>2</sup>

	Dark Resis	itance A	Spot Scan Sen	sitivity	Noise (p	k-pk)	Rise T	ime	Fall	Lime
Detector	Before	After	Before	After	Before	After	Before	After	Before	After
AD18-M7 No. 5	$1.2 \times 10^{11}$	2.5 × 10 <sup>11</sup>	50 mV +5 m(V	50 mV	4 BV	5 mV	50 ms	50 ms	70 ms	70 ms
AD18-M7 No. 6	2.3 × 10 <sup>11</sup>	4 × 10 <sup>11</sup>	80 mV +1.2 m(V	Vm 06 (	6 mV	7 mV	90 ms	90 ms	150 ms	120 ms
AD18-M7 No. 7	5.4 × 10 <sup>11</sup>	7 × 10 <sup>11</sup>	460 mS +5 m(V	440 mV	1.6 mV	78 P	2.2 s	2.2 s	150 ms	150 ms



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Figure 27 shows a typical gamma event response of a GaP detector. It is readily seen that the rise time is very fast. The delay time is about 8 ms.

Rms noise was measured as a function of gamma flux and is shown in Figure 28. There is no measureable increase in detector noise until the gamma flux is increased above  $1 \times 10^6 \ \gamma/cm^2$ -s.

Figure 29 is a graph of event rate in pulses per second, as detected by the GaP photoconductive detector, versus gamma flux. The slope of this curve indicates that the event rate of the detector is directly proportional to the gamma flux, as expected.

During the gamma radiation test, detector dark resistance was measured with a Keithley meter 602 as a funtion of gamma flux. The dark resistance was decreased by a factor of 10 when the gamma flux reached 1.5 x  $10^{10} \ \gamma/cm^2$ -s. With the same experimental setup, an Eltec resistor was measured. It changed by a factor of two under the same high gamma flux (see Figure 30). This suggests that there may be a radiation-induced current in the measuring circuit.

Three photoconductive GaP detectors (AD18L17 No. 1, AD18L17 No. 2, AD18L-9B) were exposed to a cobalt source with a total dose of  $1 \times 10^6$  rads over a 6 - 7 hour period. The dark resistance, spot scan sensitivity, rms noise, and rise time were measured before and after the gamma radiation dose for each individual detector. The results are tabulated in Table 8. One can see that there is not measureable change before and after radiation exposure and GaP is indeed a radiation-hard semiconductor material at these dose levels.

### 3.8 THERMAL VACUUM

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Eight Cu-doped GaP slit detectors were used in the thermal vacuum test which was run in a vacuum chamber. The detectors were mounted on a Honeywell built test apparatus which automatically cycles the tested detectors between  $30^{\circ}$ F and  $160^{\circ}$ F every 90 minutes (see Figure 31). The detector temperature was continuously monitored for the entire test period (151.5 hours, 101 cycles). The test chamber was maintained at a pressure of  $10^{-6}$  torr or less.

Detector dark resistance was taken before and during the thermal vacuum rest at  $30^{\circ}$ F,  $70^{\circ}$ F and  $160^{\circ}$ F. The results were tabulated in Table 9. There was no eidence of any electrical or mechanical deterioration. We believe that the slight reduction (about 10%) of detector resistance at  $160^{\circ}$ F is <u>not</u> due to the thermal effect, i.e., increase of bulk carrier concentration as temperature



Figure 27

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DETECTOR RESPONSE TO A GAMMA EVENT



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Note: The resistance change may be due to the induced current in the measurement circuit

Figure 30 DETECTOR DARK RESISTANCE VS GAMMA FLUX

### Table 8 GAMMA RADIATION EXPOSURE

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UKE	rads)
EAFUSI	x 10 <sup>6</sup>
ITON	= 1
<b>WAULA</b>	dose
- HUNH	(total

	Dark Resis	tance	Spot Scan S	ignal +2 m(V)	Noise	(rms)	Rise Ti	Lme
Detector ID	Before	After	Before	After	Before	After	Before	After
AD18-A17 No.1	$4.2 \times 10^{10}$	4.0 × 10 <sup>10</sup>	25 mV	25 mV	420 HV	440 HV	200 ms	200 ms
AD18-L17 No. 2	$1.0 \times 10^{10}$	9.0 × 10 <sup>9</sup>	120 mV	150 mV	1.0 mV	1.1 mV	80 ms	80 ms
AD18-L19 No. 3	1.4 x 10 <sup>8</sup>	1.4 × 10 <sup>8</sup>	40 mV	40 mV	3.8 aV	3.8 mV	60 ms	60 ms



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Table 9

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### DARK RESISTANCE MEASUREMENT DURING THERMAL VACUUM TEST

	Dark Resistance (ohm)	Dark Resi	stance (ohm) Du	iring Thermal Vacuum	Dark Resi	stance (ohm) Durin	g Thermal Vacuum
	Before Thermal Vacuum	at 30°F	(7-30-75) at 79°F	at 160°F (Heater on)	at 30°F	(8-4-75) at 70°F at	160°F (Heater on
AD18-M7 No. 1	7 × 10 <sup>11</sup>	6 x 10 <sup>11</sup>	6 × 10 <sup>11</sup>	1 × 10 <sup>10</sup>	6 × 10 <sup>11</sup>	3 × 10 <sup>11</sup>	8 × 10 <sup>10</sup>
AD18-M7 No. 2	2.45 × 10 <sup>11</sup>	2.9 × 10 <sup>11</sup>	3 × 10 <sup>11</sup>	$1.3 \times 10^{10}$	2.2 × 10 <sup>11</sup>	2.2 × 10 <sup>11</sup>	4 × 10 <sup>10</sup>
AD18-M7 No. 3	9 x 10 <sup>10</sup>	9 × 10 <sup>10</sup>	8.5 × 10 <sup>10</sup>	2.4 × 10 <sup>10</sup>	9 × 10 <sup>10</sup>	9 × 10 <sup>10</sup>	8.9 × 10 <sup>10</sup>
AD18-M7 No. 4	1.8 x 10 <sup>11</sup>	2.5 × 10 <sup>11</sup>	3.2 × 10 <sup>11</sup>	4 × 10 <sup>10</sup>	2 × 10 <sup>11</sup>	1.8 × 10 <sup>11</sup>	4 × 10 <sup>10</sup>
AD18-M7 No. 8	5.4 × 10 <sup>9</sup>	3.9 × 10 <sup>9</sup>	3.6 × 10 <sup>9</sup>	1 × 10 <sup>9</sup>	1.7 × 10 <sup>9</sup>	1.7 × 10 <sup>9</sup>	7 × 10 <sup>8</sup>
AD18-M7 No. 9	3.2 × 10 <sup>11</sup>	4.1 × 10 <sup>11</sup>	4.9 x 10 <sup>11</sup>	3.5 × 10 <sup>10</sup>	6.5 × 10 <sup>11</sup>	$4.1 \times 10^{11}$	$3.7 \times 10^{11}$
AD24-L1 No. 1	8 × 10 <sup>5</sup>	8 × 10 <sup>5</sup>	7.1 × 10 <sup>5</sup>	7.3 × 10 <sup>5</sup>	8 × 10 <sup>5</sup>	7.7 × 10 <sup>5</sup>	7.4 × 10 <sup>5</sup>
AD24-L1 No. 4	9.4 × 10 <sup>8</sup>	9 × 10 <sub>6</sub>	8.6 x 10 <sup>6</sup>	8.6 × 10 <sup>6</sup>	9 × 10 <sub>6</sub>	8.6 x 10 <sup>6</sup>	8.5 × 10 <sup>6</sup>

Table 9 (Cont.)

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### DARK RESISTANCE MEASUREMENT AT THERMAL VACUUM TEST

			Dark Resi Vaci	stance During Thermal uum (8-5-75)	Dark Resistance After Thermal Vacuum
			at 30°F	at 70°F at 160°F	
AD18-M7	No.	Ч	$6 \times 10^{11}$	$1.6 \times 10^{11} 1 \times 10^{11}$	$6 \times 10^{11}$
AD18-M7	No.	2	$3 \times 10^{11}$	$2 \times 10^{11} 4.5 \times 10^{10}$	$2.4 \times 10^{11}$
AD18-M7	No.	3	$9.4 \times 10^{10}$	$9 \times 10^{10} 8 \times 10^{10}$	$9 \times 10^{10}$
AD18-M7	No.	4	$3.5 \times 10^{11}$	$2.4 \times 10^{11} 3.5 \times 10^{10}$	$3.2 \times 10^{11}$
AD18-M7	No.	8	$1.4 \times 10^{9}$	$1 \times 10^9$ $7 \times 10^8$	$2.8 \times 10^9$
AD18-M7	No.	6	$6.5 \times 10^{11}$	$5.1 \times 10^{11} 3.4 \times 10^{11}$	$5.7 \times 10^{11}$
AD24-L1	No.	٦	$8.2 \times 10^{5}$	$8.1 \times 10^5$ 7.9 × $10^5$	7 x 10 <sup>5</sup>
AD24-L1	No.	4	$9 \times 10^{6}$	$8.5 \times 10^6 8.5 \times 10^6$	8.6 x 10 <sup>6</sup>

is increased. The heating apparatus includes an electric heater. While the thermal cycle was at 160°F, the heater was on and glowed red. The glowing heater acted as an optical light source and generated free electrons in the detector and reduced the detector resistance.

An initial electrical test was performed. Because the amplifier was not kept inside the chamber during the thermal vacuum cycling, the detector output was connected to the input terminal of the amplifier through long (about 10 ft) shielded cables. Excess noise was produced in the cables. Therefore, for this test only relative signal data before, during and after the thermal vacuum were reported.

A small light bulb was placed inside the shield dark chamber. Signals through the amplifier were taken before, after, and during the thermal vacuum cycle at 30°F, 70°F and 160°F (see Figure 32). One can easily see that there is no performance variation at different temperature points.

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Detector AD18M7 No. 9 Signal before thermal vacuum cycle



Signal during thermal vacuum cycle at 30°F



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Signal during thermal vacuum cycle at 75°F

Figure 32 DETECTOR SIGNAL DURING THERMAL VACUUM TEST 62


Detector AD18M7 No. 9 Signal during thermal vacuum cycle at 160°F

Signal after thermal vacuum cycle at room temperature

# SECTION IV

Cu-doped GaP photoconductive slit detectors with the dimensions 0.05 x 0.0005 inch<sup>2</sup> have been developed for star sensor applications. The sensitivity of these devices approaches the performance of photomultipliers, and their response times are faster than those of CdS and CdSe. For example, noise equivalent powers as low as 6 x  $10^{-16}$  W/VHz at 0.5 µm and corner frequencies as high as 50 Hz have been observed on these devices. The GaP photoconductive detectors are environmentally stable and exhibit high radiation resistance. Test results obtained on this program. indicated that Cu-coped GaP slit detectors exhibit no permanent degradation after exposure to 200°F; 90% ± 4% relative humidity; vibration from 400 to 3,000 Hz under 25 g; electron irradiation to 10<sup>15</sup> e/cm<sup>2</sup>; proton irradiation to 10<sup>11</sup> p/cm<sup>2</sup>: and gamma ray exposure to 10<sup>b</sup> rads (Si). The lack of spot scan uniformity along the slits is the chief drawback of these devices. Therefore, for the photoconductive mode Cu-doped GaP detectors, further material growth development effort is required to achieve accurate compensation, reproducibility and uniformity.

設計

### APPENDIX

## CALCULATION OF DARK RESISTANCE OF A SLIT DETECTOR WITH PARALLEL PLATE ELECTRODE CONFIGURATION

A parallel plate electrode configuration sketched below is used to make contact to the slit detectors, where  $\lambda$ , W are the slit width any length, respectively, and d is the width of the electrodes.



The dark resistance, when there is no incident light flux, is equivalent to the volume shunt which can be calculated from electrostatic theory.(1)

$$R_{d} = \frac{\kappa \left(\frac{\ell}{2d+\ell}\right)}{\sigma_{d} W K \left(\frac{2d^{1/2} (d+\ell)^{1/2}}{2d+\ell}\right)}$$
(1)

where  $\sigma_d$  is the bulk conductivity of the material and K(k) is a complete elliptic integral of modules k. For the present program, the slit detector has dimensions l, d, W = 0.5, 3.5, 50 mils, respectively, the calculated dark resistance from equation lis

$$R_{d} = \frac{2.8}{\sigma_{d}}$$
(2)

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