

# AADL and Model-based Engineering

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## Report Documentation Page

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# Outline

- ▶ Challenges in Safety-critical Software-intensive systems
- An Architecture-centric Virtual Integration Strategy with SAE AADL
- Improving the Quality of Requirements
- Architecture Fault Modeling and Safety
- Incremental Life-cycle Assurance of Systems
- Summary and Conclusion



# We Rely on Software for Safe Aircraft Operation

## Quantas Landing

Written by htbw  
From: soyawan



Even with the autopilot off, flight control computers still `` command control surfaces to protect the aircraft from unsafe conditions such as a stall," the investigators said.

The unit continued to send false stall and speed warnings to the aircraft's primary computer and about 2 minutes after the initial fault `` generated very high, random and incorrect values for the aircraft's angle of attack."

mayday call when it suddenly changed altitude during a flight

from Singapore to Perth, Qantas said.

**Embedded software systems introduce a new class of problems not addressed by traditional system modeling & analysis**

plunge

Autopilot Off

A `` preliminary analysis" of the Qantas plunge showed the error occurred in one of the jet's three air data inertial reference units, which caused the autopilot to disconnect, the ATSB said in a statement on its Web site.

The crew flew the aircraft manually to the end of the flight, except for a period of a few seconds, the bureau said.

Even with the autopilot off, flight control computers still `` command control surfaces to protect the aircraft from unsafe conditions such as a stall," the investigators said.

The unit continued to send false stall and speed warnings to the aircraft's primary computer and about 2 minutes after the initial fault `` generated very high, random and incorrect values for the aircraft's angle of attack."

The flight control computer then commanded a nose-down aircraft movement, which resulted in the aircraft pitching down to a maximum of about 8.5 degrees," it said.

No `` Similar Event'

`` Airbus has advised that it is not aware of any similar event over the many years of operation of the Airbus," the bureau added, saying it will continue investigating.

Let's flight simulator on the autopilot and generated false data, causing the jet to nosedive.

was cruising at 37,000 feet (11,277 meters) when the computer fed incorrect information to the flight control system, the **Australian Transport Safety Bureau** said yesterday. The aircraft dropped 650 feet within seconds, slamming passengers and crew into the cabin ceiling, before the pilots regained control.

`` This appears to be a unique event," the bureau said, adding that

fitted with the same air-data computer. The advisory is `` aimed at minimizing the risk in the unlikely event of a similar occurrence."



# Software Problems not just in Aircraft



Expert • Independent • Nonprofit  
**ConsumerReports.org**<sup>®</sup>



This article appeared in [May 2010 Consumer Reports Magazine](#). But it turned out to be a problem for the Kenmore 4027 front-loader, which scored near the bottom in our [February 2010 report](#).

May 7, 2010

## Lexus GX 460 passes retest; Consumer Reports lifts "Don't Buy" label

Consumer Reports is lifting the [Don't Buy: Safety Risk](#) designation from the [2010 Lexus GX 460 SUV](#) after recall work corrected the problem it displayed in one of our emergency handling tests. (See the original report and video: "[Don't Buy: Safety Risk--2010 Lexus GX 460](#).")



We originally experienced the problem in a test that we use to evaluate what's called lift-off oversteer. In this test, as the vehicle is driven through a turn, the driver quickly lifts his foot off the accelerator pedal to see how the vehicle reacts. When we did this with our GX 460, its rear end slid out until the vehicle was almost sideways. Although the GX 460 has [electronic stability control](#), which is designed to prevent a vehicle from sliding, the system wasn't intervening quickly

enough to stop the slide. We consider this a safety risk because in a real-world situation this could cause a rear tire to strike a curb or slide off of the pavement, possibly causing the vehicle to roll over. Tall vehicles with a high center of gravity, such as the GX 460, heighten our concern. We are not aware, however, of any reports of injury related to this problem.

Lexus recently duplicated the problem on its own test track and developed a [software upgrade](#) for the vehicle's ESC system that would prevent the problem from happening. [Dealers received the software fix](#) last week and began notifying GX 460 owners to bring their vehicles in for repair.

We contacted the Lexus dealership from which we had anonymously bought the vehicle and made an appointment to have the recall work performed. The work took about an hour and a half.

Following that, we again put the SUV through our full series of emergency handling tests. This time, the ESC system intervened earlier and its rear did not slide out in the lift-off oversteer test. Instead, the vehicle understeered—or plowed—when it exceeded its limits of traction, which is a more common result and makes the vehicle more predictable and less likely to roll over. Overall, we did not experience any safety concerns with the corrected GX 460 in our handling tests.

## "Don't Buy"

Many appliances now rely on electronic controls and operating software. But it turned out to be a problem for the Kenmore 4027 front-loader, which scored near the bottom in our [February 2010 report](#).

Our tests found that the rinse cycles on some models worked improperly, resulting in an unimpressive cleaning.

When Sears, which sells the washer, saw our [February 2010 Ratings](#) (available to subscribers), it worked with LG, which makes the washer, to figure out what was wrong. They quickly determined that a software problem was causing short or missing rinse and wash cycles, affecting wash performance. Sears and LG say they have reprogrammed the software on the models in their warehouses and on about 65 percent of the washers already sold, including the ones we had purchased.

Our retests of the reprogrammed Kenmore 4027 found that the cycles now worked properly, and the machine excelled. It now tops our [Ratings](#) (available to subscribers) of more than 50 front-loaders and we've made it a CR Best Buy.

If you own the washer, or a related model such as the Kenmore 4044 or Kenmore Elite 4051 or 4219, you should get a letter from Sears for a free service call. Or you can call 800-733-2299.

**How do you upgrade washing machine software?**



# High Fault Leakage Drives Major Increase in Rework Cost

*Aircraft industry has reached limits of affordability due to exponential growth in SW size and complexity.*

**70% Requirements & system interaction errors**

**80% late error discovery at high rework cost**

**20.5% 300-1000x**

**0%, 9% 80x**

**70%, 3.5% 1x**

**10%, 50.5% 20x**

**Major cost savings through rework avoidance by early discovery and correction**

**A \$10k architecture phase correction saves \$3M**

**20%, 16% 5x**

**Total System Cost  
Boeing 777 \$12B  
Boeing 787 \$24B**

**Software as % of total system cost  
1997: 45% → 2010: 66% → 2024: 88%**

**Post-unit test software rework cost  
50% of total system cost and growing**

*Where faults are introduced*

*Where faults are found*

*The estimated nominal cost for fault removal*

Sources:

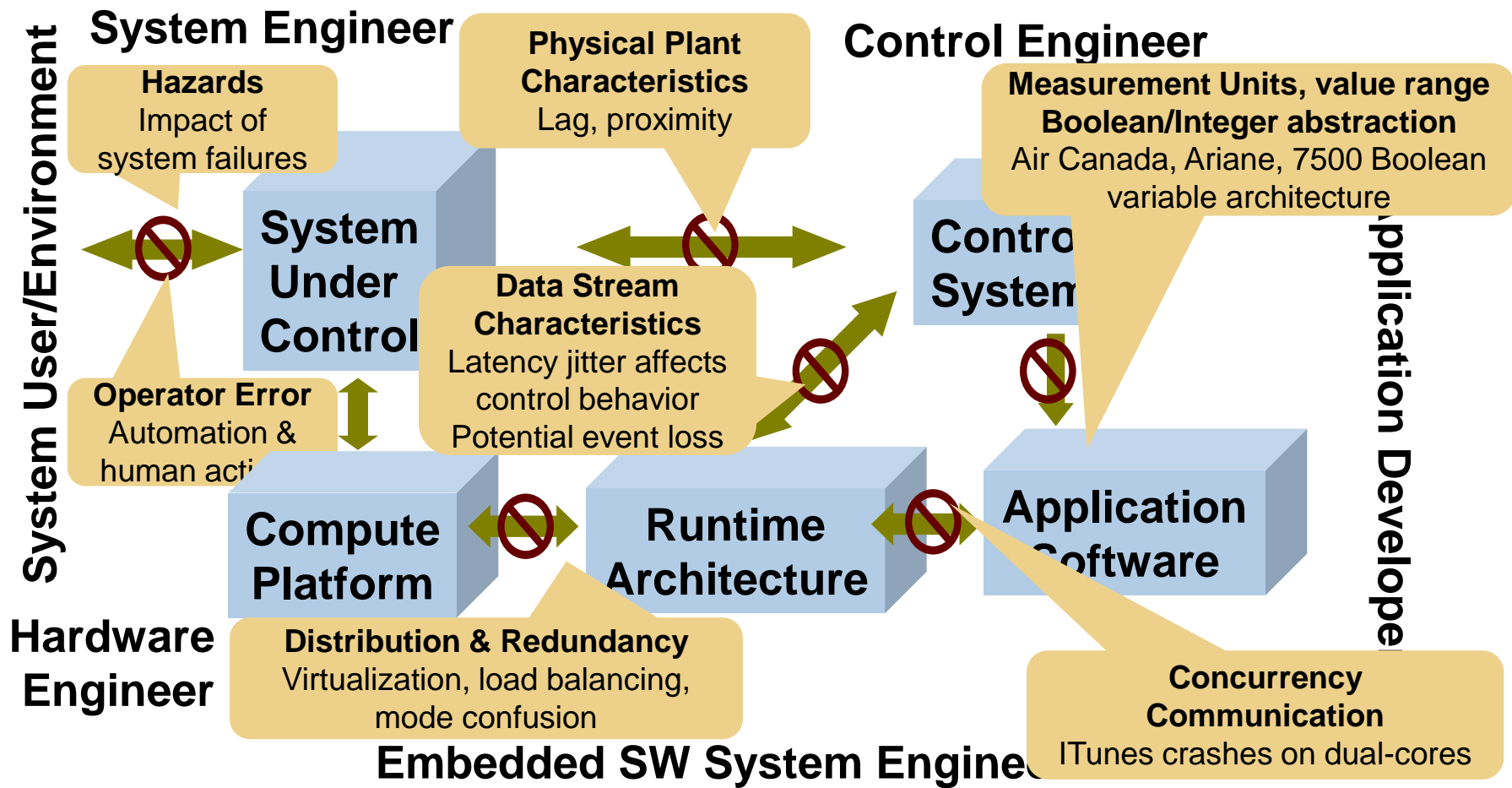
NIST Planning report 02-3, *The Economic Impacts of Inadequate Infrastructure for Software Testing*, May 2002.

D. Galin, *Software Quality Assurance: From Theory to Implementation*, Pearson/Addison-Wesley (2004)

B.W. Boehm, *Software Engineering Economics*, Prentice Hall (1981)



# Mismatched Assumptions in System Interactions



*Embedded software system as major source of hazards*

*Why do system level failures still occur despite fault tolerance techniques being deployed in systems?*



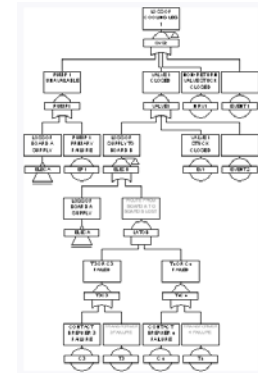
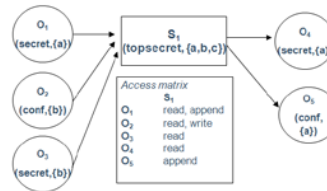
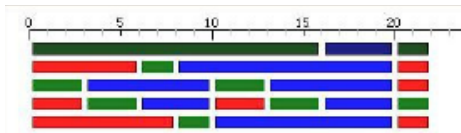


# Model-based Engineering Pitfalls



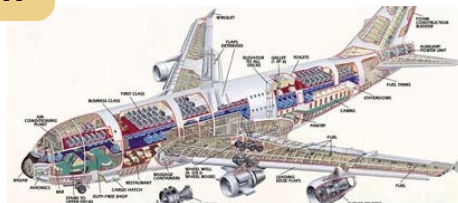
The system

Inconsistency between independently developed analytical models



System models

Confidence that model reflects implementation



System implementation

This aircraft industry experience has led to the System Architecture Virtual Integration (SAVI) initiative

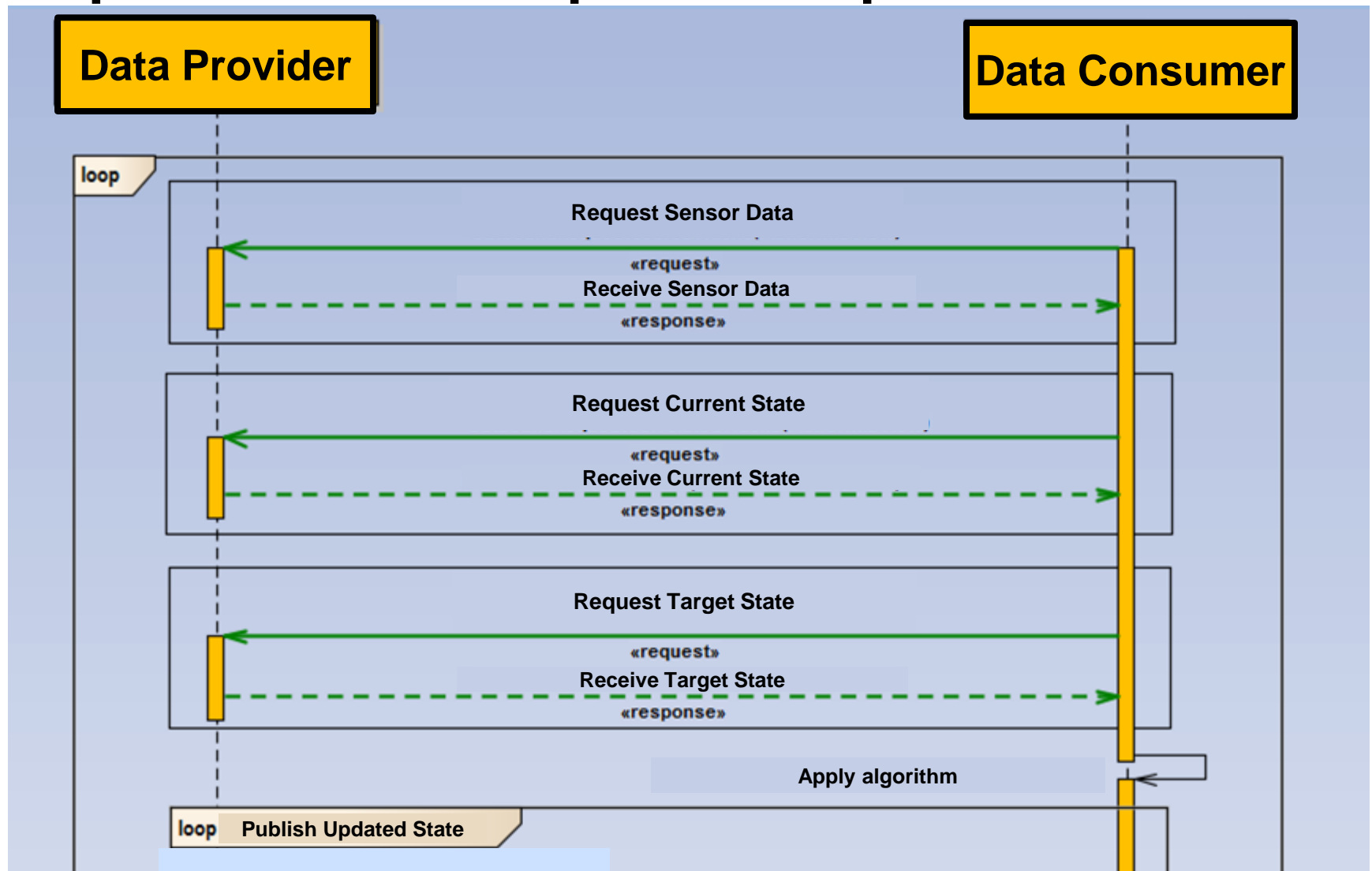


# Why UML, SysML Are Not Sufficient

- System engineering
  - Focus on system architecture and operational environment
  - SysML developed to capture interactions with outside world, as a standardized UML profile
  - 4 pillars/diagrams: requirements, parameterics (added in SysML), structure, behavior
- Conceptual architecture
  - UML-based component model
  - Architecture views (DoDAF, IEEE 1471)
  - Platform Independent model (PIM)
- Embedded software system engineering
  - OMG Modeling and Analysis of Real Time Embedded systems (MARTE) as UML profile
    - Borrowed Meta model concepts from AADL
    - Focus on modeling implementations
  - xUML insufficient for PSM (Kennedy-Carter, NATO ALWI study)



# Impact of Three Step Data Request Protocol



# Operating as ARINC653 Partitioned System

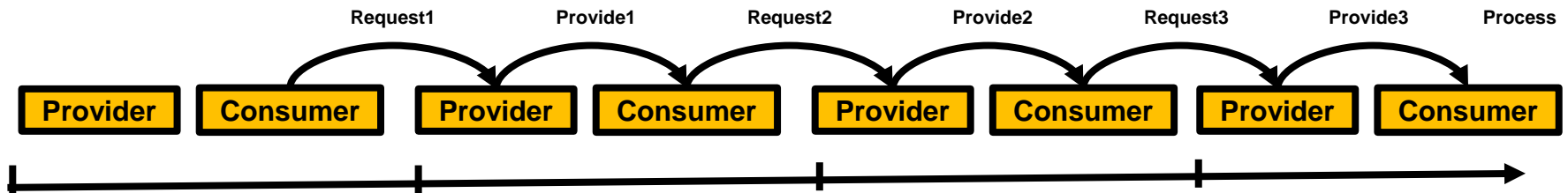
## Data Consumer Requirement

- Process data in 1 second

## Partitions

- Provide space and time boundary enforcement
- Execute periodically on a static timeline at 1 second rate

## Data request protocols across partitions



**How much time does consumer actually have to process the data?  
Who pays for the communication overhead?**



# Model-based Engineering in Practice

## Modeling is used in practice

- Modeling, analysis, and simulation in mechanical, control, computer hardware engineering

## Current practice: modeling and software

- Remember software through pictures
- MDE and MDA with UML
- Automatically generated documents

## We need language for architecture modeling

- Strongly typed
- Well-defined execution and communication timing semantics
- Systematic approach to dealing with exceptional conditions
- Support for large-scale development



# Outline

Challenges in Safety-critical Software-intensive systems

▶ An Architecture-centric Virtual Integration Strategy with SAE AADL

Improving the Quality of Requirements

Architecture Fault Modeling and Safety

Incremental Life-cycle Assurance of Systems

Summary and Conclusion



# The Roots of AADL

1990-1998: MetaH and Control-H by Steve Vestal

- Strong typing, syntax borrowed from Ada
- Data and event ports, Operational modes
- Scheduling analysis and code generation

1994: Application to Missile Guidance System by Vestal and Lewis

- Three Week Port to Dual Processor Hardware

1997: MetaH Style for ACME by Peter Feiler and Jun Li

- CMU ECE Ph.D. on multi-dimensional analysis for Simplex architectures

1998: Error Model added to MetaH by Steve Vestal

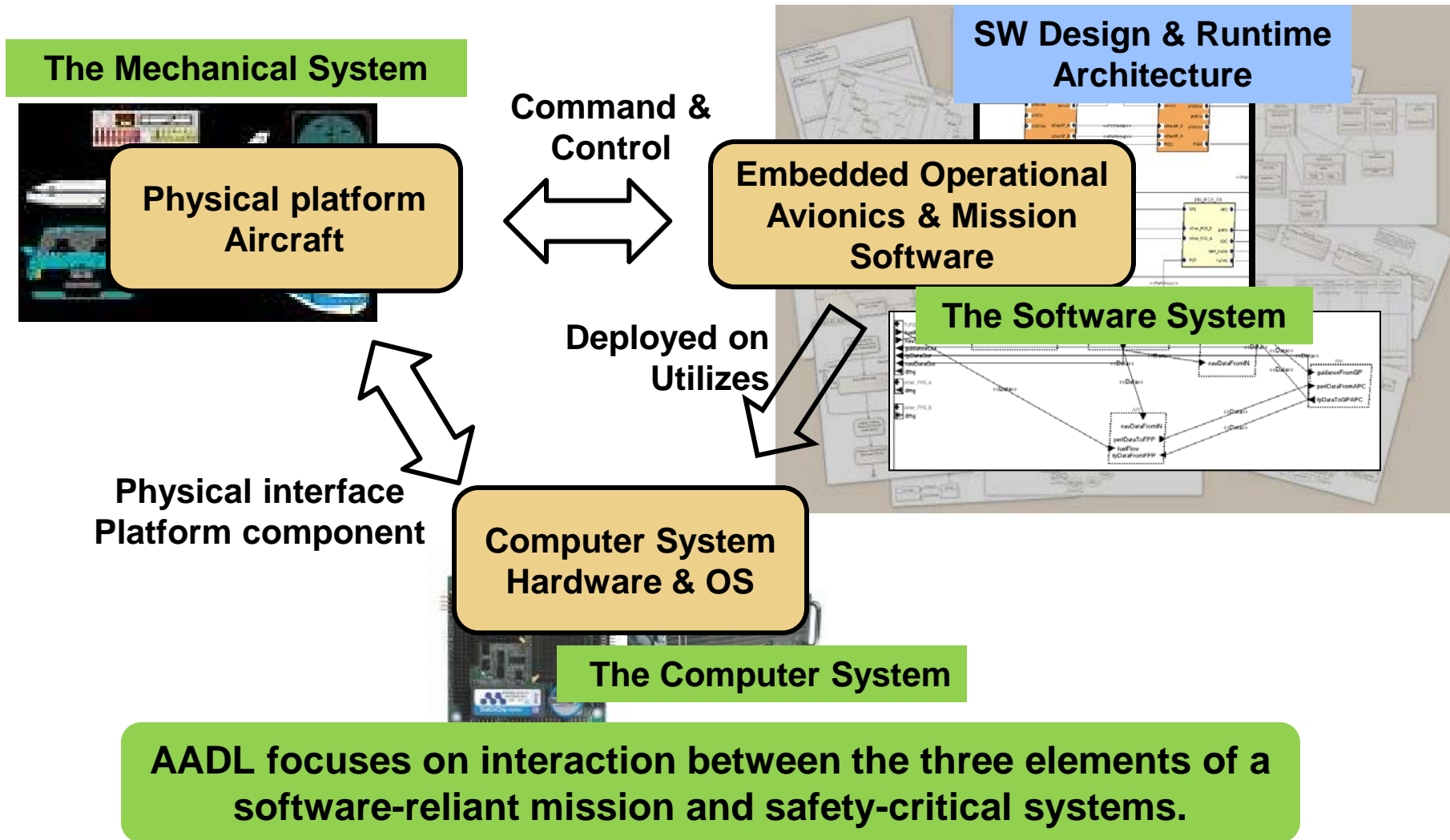
- Generation of fault trees and Markov models

1999: Requirements Document for AADL Standard

- Industry input: packages, messages
- The best of MetaH and ACME



# SAE Architecture Analysis & Design Language (AADL) for Software-reliant Systems





# The SAE AADL Standard Suite (AS-5506 series)

Core AADL language standard (V2.1-Sep 2012, V1-Nov 2004)

- Strongly typed language with well-defined semantics
- Textual and graphical notation
- Standardized XMI interchange format

## Standardized AADL Extensions

Error Model language for safety, reliability, security analysis

ARINC653 extension for partitioned architectures

Behavior Specification Language for modes and interaction behavior

Data Modeling extension for interfacing with data models (UML, ASN.1, ...)

## AADL Annex Extensions in Progress

Requirements Definition and Assurance Annex

Synchronous System Specification Annex

Hybrid System Specification Annex

System Constraint Specification Annex

Network Specification Annex



# System Level Fault Root Causes

## Violation of data stream assumptions

- Stream miss rates, Mismatched data representation, Latency jitter & age

**End-to-end latency analysis  
Port connection consistency**

## Partitions as Isolation Regions

- Space, time, and bandwidth partitioning
- Isolation not guaranteed due to undocumented resource sharing
- fault containment, security levels, safety levels, distribution

**Process and virtual processor to  
model partitioned architectures**

## Virtualization of time & resources

- Logical vs. physical redundancy
- Time stamping of data & asynchronous systems

**Virtual processors & buses  
Multiple time domains**

## Inconsistent System States & Interactions

- Modal systems with modal components
- Concurrency & redundancy management
- Application level interaction protocols

**Operational and failure modes  
Interaction behavior specification  
Dynamic reconfiguration  
Fault detection, isolation, recovery**

## Performance impedance mismatches

- Processor, memory & network resources
- Compositional & replacement performance mismatches
- Unmanaged computer system resources

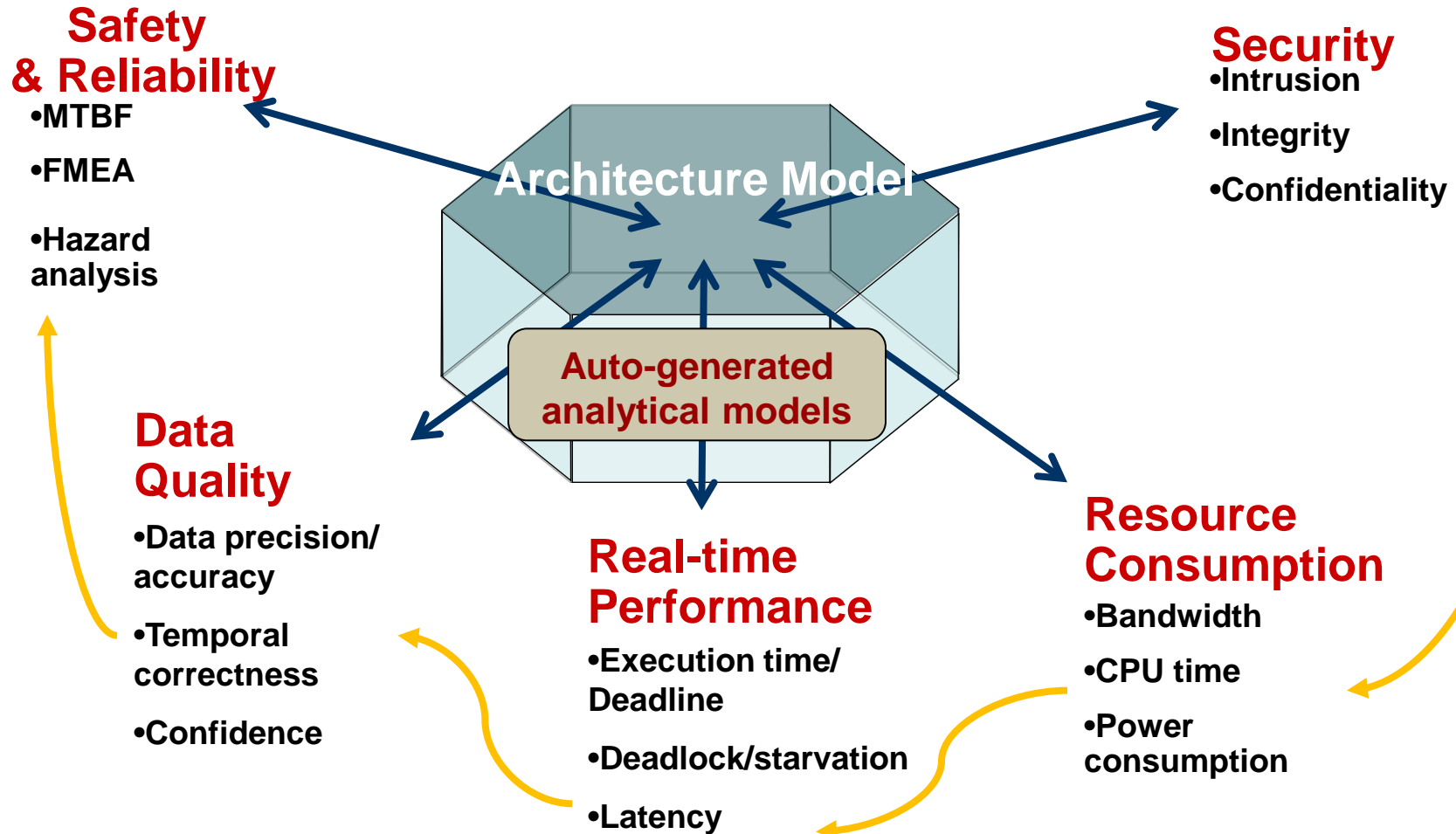
**Resource allocation &  
deployment configurations  
Resource budget analysis  
& scheduling analysis**

Codified in Virtual Upgrade Validation method

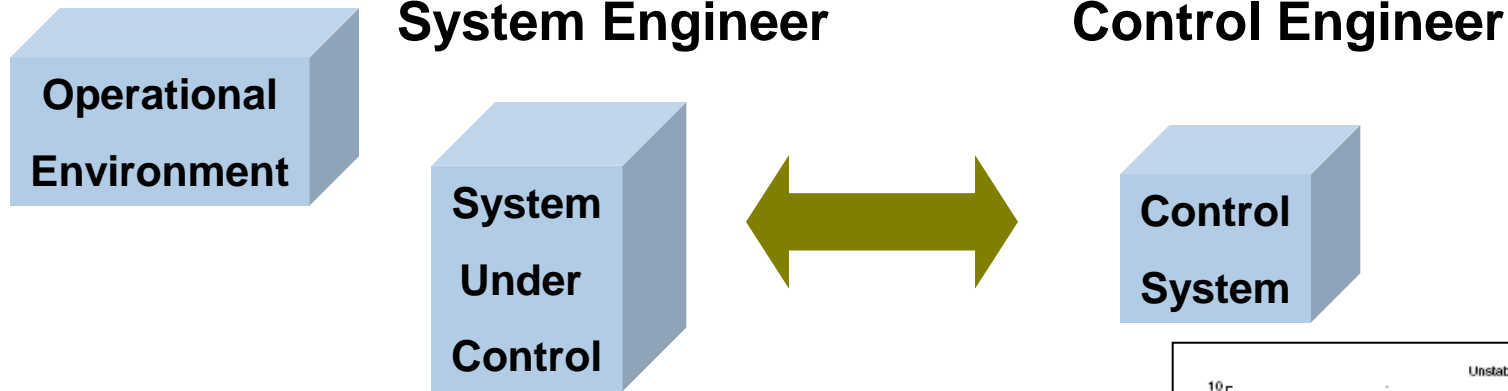


# Architecture-Centric Quality Attribute Analysis

Single Annotated Architecture Model Addresses Impact Across Operational Quality Attributes

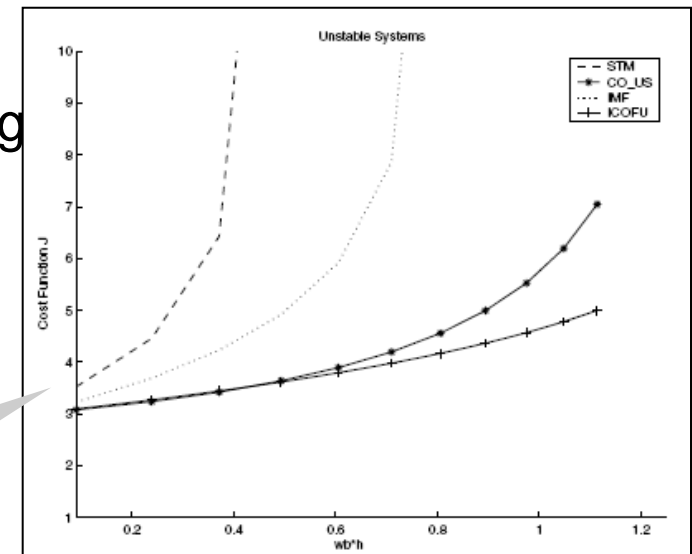


# Multi-Fidelity End-to-end Latency in Control Systems



Common latency data from system engineering

- Processing latency
- Sampling latency
- Physical signal latency



Impact of Scheduler Choice on Controller Stability

A. Cervin, Lund U., CCACSD 2006



# Software-Based Latency Contributors

Execution time variation: algorithm, use of cache

Processor speed

Resource contention

Preemption

Legacy & shared variable communication

Rate group optimization

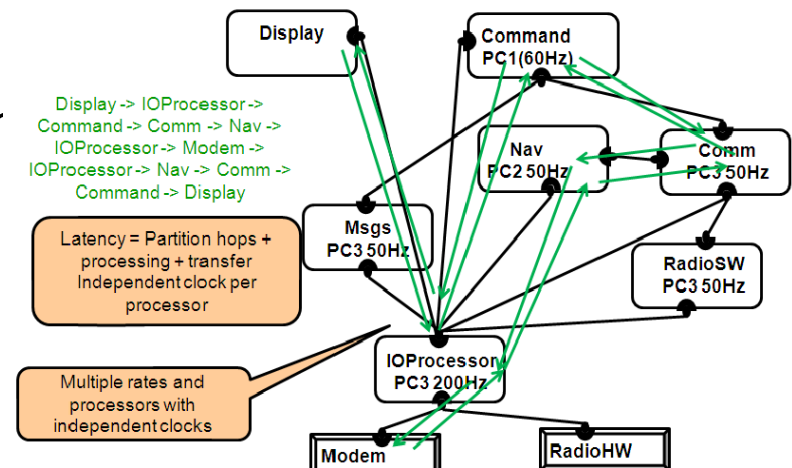
Protocol specific communication delay

Partitioned architecture

Migration of functionality

Fault tolerance strategy

Flow Use Scenario through Subsystem Architecture



# Sampling of International Efforts Leveraging SAE AADL

Compositional  
Timing Framework  
OSD 2014



P Project  
Auto Code Gen  
2011-2014



OPEES  
Formal analysis  
2011-2014



OpenGroup  
Real-Time Forum  
EU + US partners  
2008-current



D-MILS  
Design of Secure Systems  
2013 - \$4.9M



ESA COMPASS  
System SW Co-engineering  
2008-current

TOPCASED  
Open Source Embedded  
Systems Tool Framework  
28 partners €20+M 2005-2009



IST ARTIST2  
Embedded Systems  
Center of Excellence  
2007-2012

AVSI SAVI  
Analysis-based System Validation  
12 partners \$20M 2008-current



ITEA SPICES  
Model-Driven Embedded  
Systems Engineering  
15 partners €16M 2006-2009



Flex-eWare  
Auto Code Generation  
2007-2010

DARPA META  
Complex System  
Engineering  
2010-2012



EC ASSERT  
Proof-based Satellite  
Architectures  
ESA + 30 partners  
€15M 2004-2009



ESA TASTE  
System & SW  
Validation & Generation  
2010-current

PARSEC  
Safety/security  
2010-2013



DARPA HACMS  
Security in CPS  
RC formal methods  
2013-2015



AADL Inspector  
Ellidiss  
2010-current

Integrated Clinical  
Environment  
Device Certification  
FDA KSU  
2011-current



PROARTIS  
Partitioned RT systems  
2010-2013 €1.8M



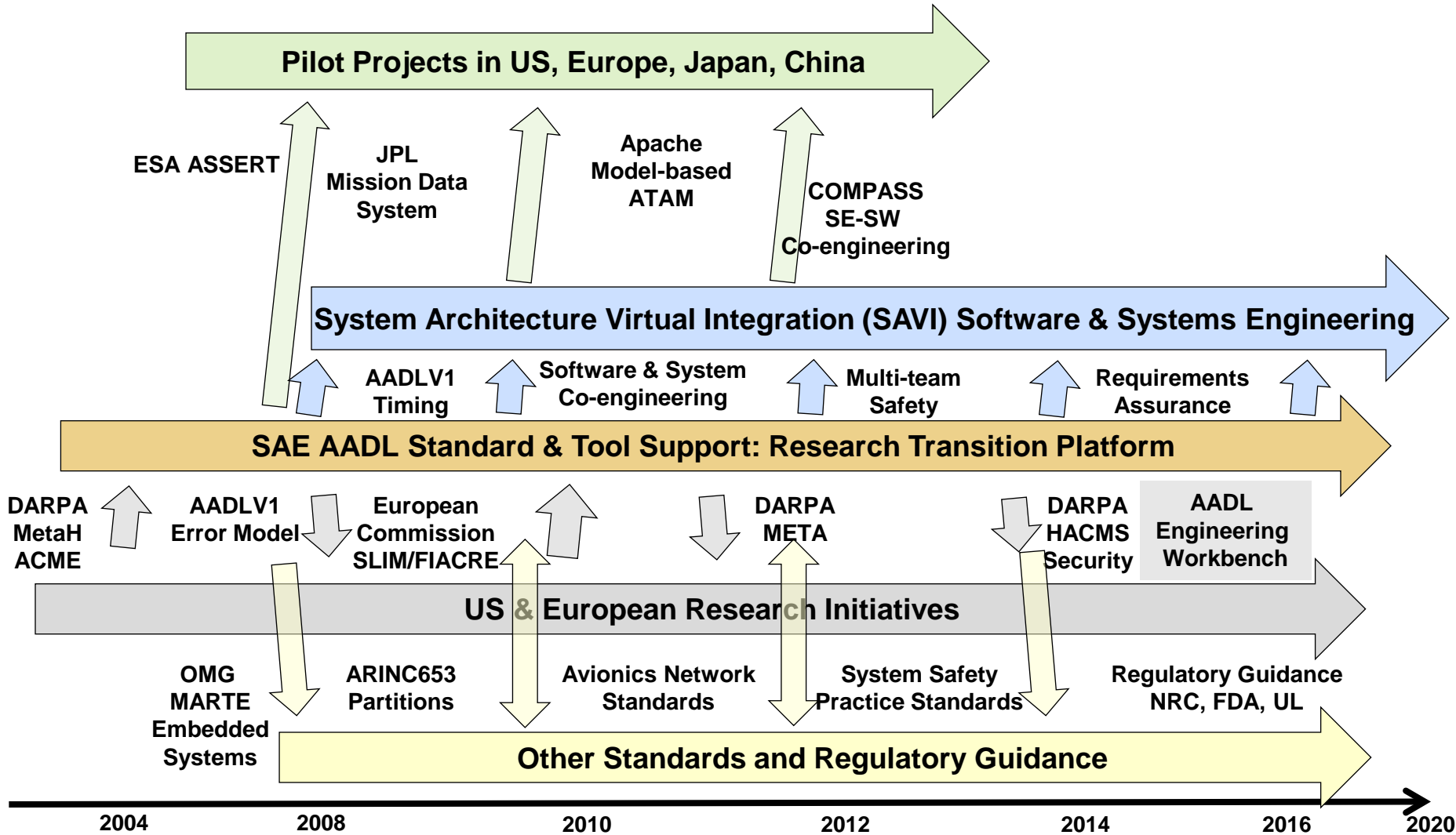
RAMSES  
Auto Code Generation  
2012-current

MASIW  
Avionics Workbench  
2011-current  
\$2M per year



# Architecture-centric Virtual System Integration

## Evolution, Maturation and Transition



# Early Discovery and Incremental V&V through System Architecture Virtual Integration (SAVI)

**Aircraft: (Tier 0)**

**Aircraft system: (Tier 1)**  
 Engine, Landing Gear, Cockpit, ...  
 Weight, Electrical, Fuel, Hydraulics,...

**LRU/IMA System: (Tier 2)**  
 Hardware platform, software partitions  
 Power, MIPS, RAM capacity & budgets  
 End-to-end flow latency

**System & SW Engineering:**  
 Mechatronics: Actuator & Wings  
 Safety Analysis (FHA, FMEA)  
 Reliability Analysis (MTTF)

**Subcontracted software subsystem: (Tier 3)**  
 Tasks, periods, execution time  
 Software allocation, schedulability  
 Generated executables

**OEM & Subcontractor:**  
 Subsystem proposal validation  
 Functional integration consistency  
 Data bus protocol mappings

**Repeated Virtual Integration Analyses:**  
 Power/weight  
 MIPS/RAM, Scheduling  
 End-to-end latency  
 Network bandwidth

## *Proof of Concept Demonstration and Transition by Aerospace industry initiative*

- Architecture-centric model-based software and system engineering
- Architecture-centric model-based acquisition and development process
- Multi notation, multi team model repository & standardized model interchange

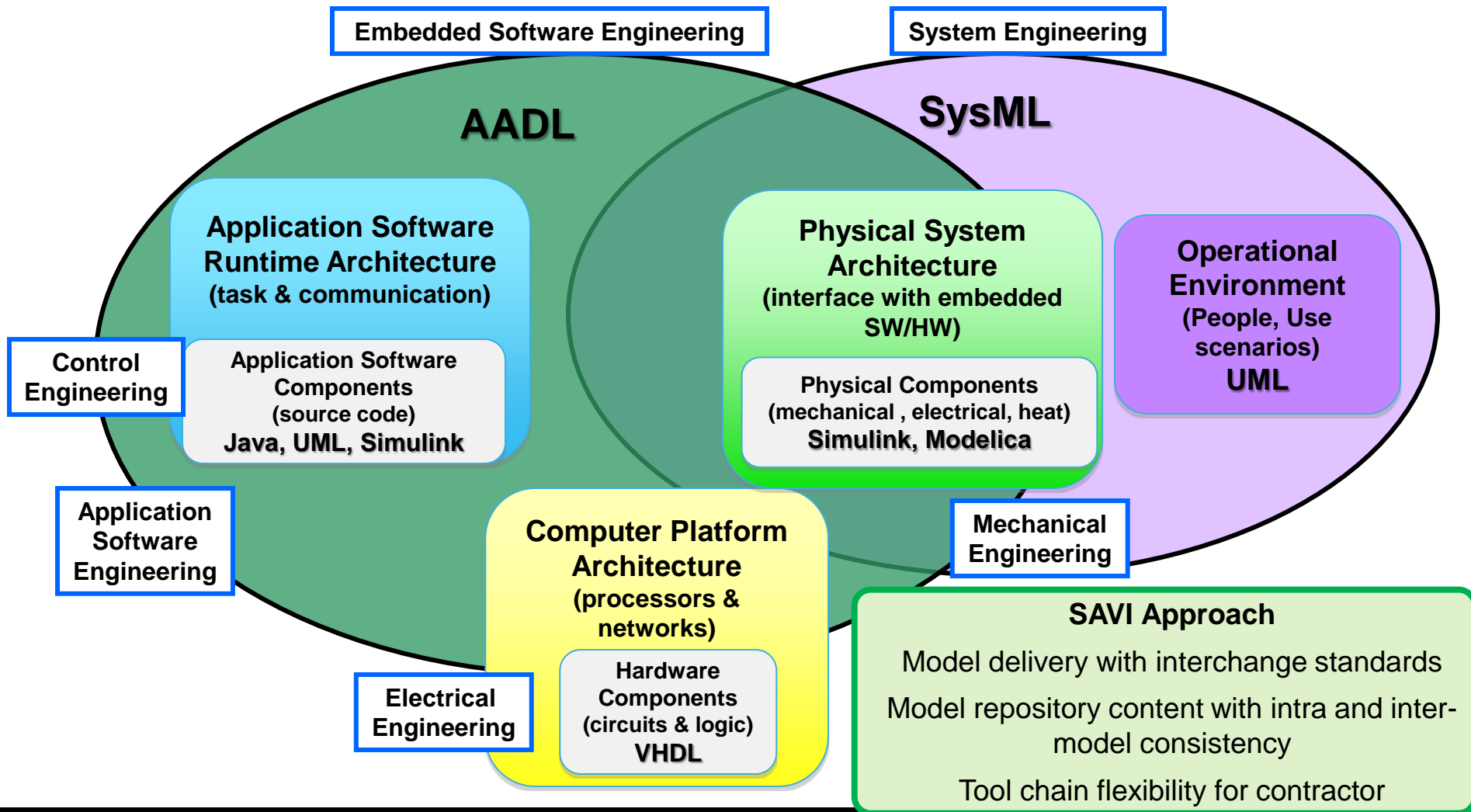
■ Multi-tier system & software architecture (in AADL)

■ Incremental end-to-end validation of system properties

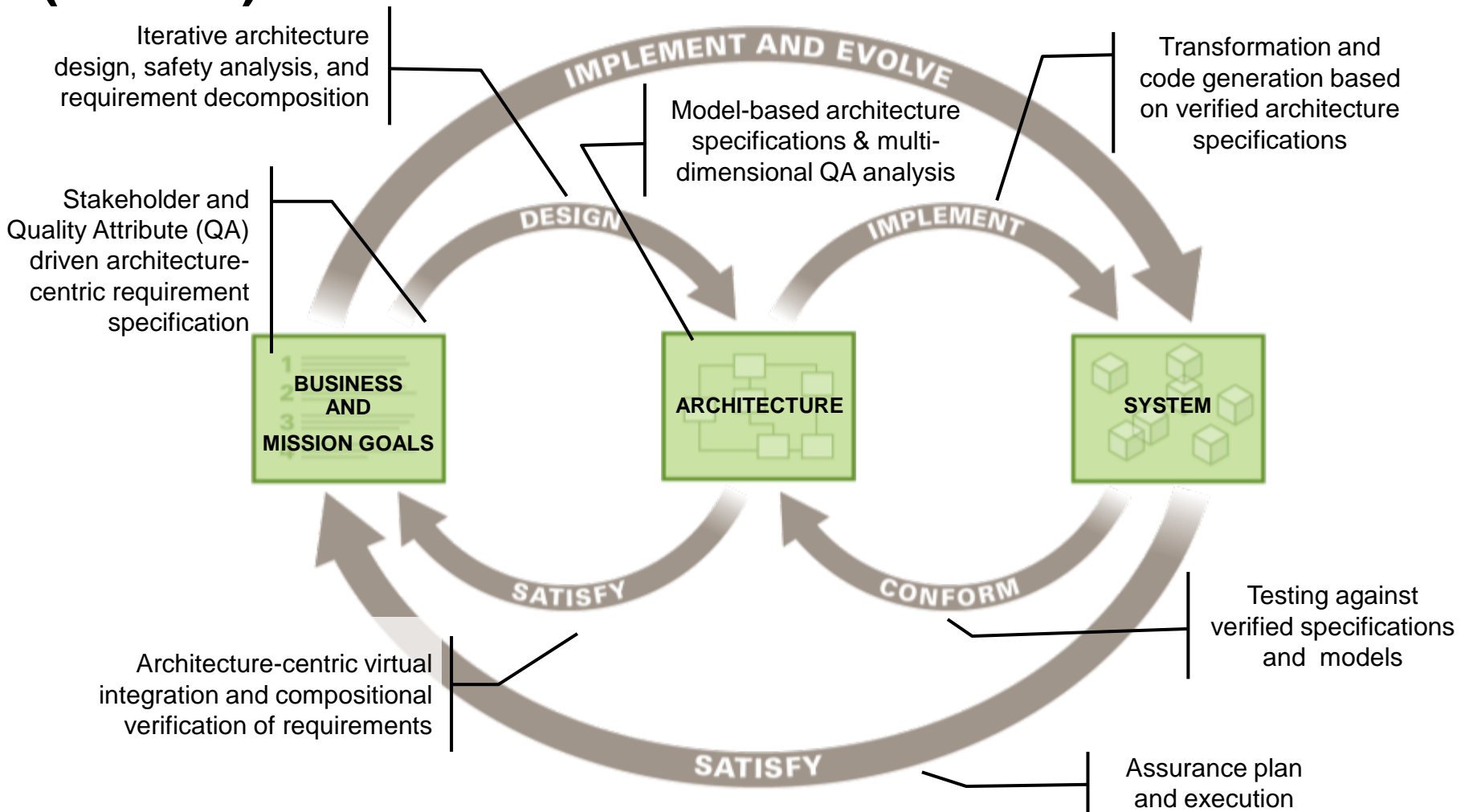




# Multi-Notation Approach to Architecture-centric Virtual System and Software Integration



# Architecture-centric Virtual Integration Practice (ACVIP)



# Outline

Challenges in Safety-critical Software-intensive systems

An Architecture-centric Virtual Integration Strategy with SAE AADL

▶ Improving the Quality of Requirements

Architecture Fault Modeling and Safety

Incremental Life-cycle Assurance of Systems

Summary and Conclusion



# Certification & Recertification Challenges

Certification: assure the quality of the delivered system

- Sufficient evidence that a system implementation meets system requirements
- Quality of requirements and quality of evidence determines quality of system

Certification related rework cost

- Currently 50% of total system cost and growing

Recertification Challenge

- Desired cost of recertification in proportion to change

Improve quality of requirements and evidence

Perform verification compositionally  
throughout the life cycle



# Current Industry Practice in DO-178B Compliant Requirements Capture

## Industry Survey in 2009 FAA Requirements Engineering Study

### Notation

Enter an "x" in every row/column cell that applies

	System Requirements	Data Interconnect {ICD}	High-Level Software Requirements	Low-Level Software Requirements	Hardware Requirements
English Text or Shall Statements	39	27	36	32	29
Tables and Diagrams	31	30	30	19	18
UML Use Cases	1		2	4	
UML Sequence Diagrams			3	6	
UML State Diagrams			1	7	
Executable Models (e.g. Simulink, SCADE Suite, etc.)	7	1	8	8	1
Data Flow Diagrams (e.g. Yourdon)	4		6	9	
Other (Specify)		1			
Operational models or prototypes	1	1			1
UML			1	1	

### Tool

Enter an "x" in every row/column cell that applies

	System Requirements	Data Interconnect {ICD}	High-Level Software Requirements	Low-Level Software Requirements	Hardware Requirements
Database (e.g. Microsoft Access)	3	4	3	3	
DOORS	23	13	22	18	12
Rational ROSE®			1	3	
RDD-100®					
Requisite Pro®	5	3	5	4	4
Rhapsody	1				
SCADE Suite	2		3	1	
Simulink	5	1	5	3	1
Slate	1		1	1	
Spreadsheet (e.g., Microsoft Excel)	5	4	5	4	3
Statemate					
Word Processor (e.g., Microsoft Word)	19	20	18	17	16
VAPSTM		1	3	3	
Designer's Workbench™			1	1	
Proprietary Database, SCADE like pic tool		1	1		
Interleaf	1	1	1	1	1
BEACON	1	1	1	1	
CaliberRM	1	1	1	1	1
XM:		1			
Wiring diagram		1			1

Need analyzable & executable specifications



# Requirement Quality Challenge

Requirements error	%
Incomplete	21%
Missing	33%
Incorrect	24%
Ambiguous	6%
Inconsistent	5%

There is more to requirements quality than “shall”’s and stakeholder traceability  
*IEEE 830-1998 Recommended Practice for SW Requirements Specification*



IEEE Std 830-1998 characteristics of a good requirements specification:

- Correct
- Unambiguous
- Complete
- Consistent
- Ranked for importance and/or stability
- Verifiable
- Modifiable
- Traceable

**System to SW requirements gap [Boehm 2006]**

*How do we verify low level SW requirements against system requirements?*

When StartUpComplete is TRUE in both FADECs and SlowStartupComplete is FALSE, the FADECStartupSW shall set SlowStartupIncomplete to TRUE



# Mixture of Requirements & Architecture Design Constraints

## Requirements for a Patient Therapy System

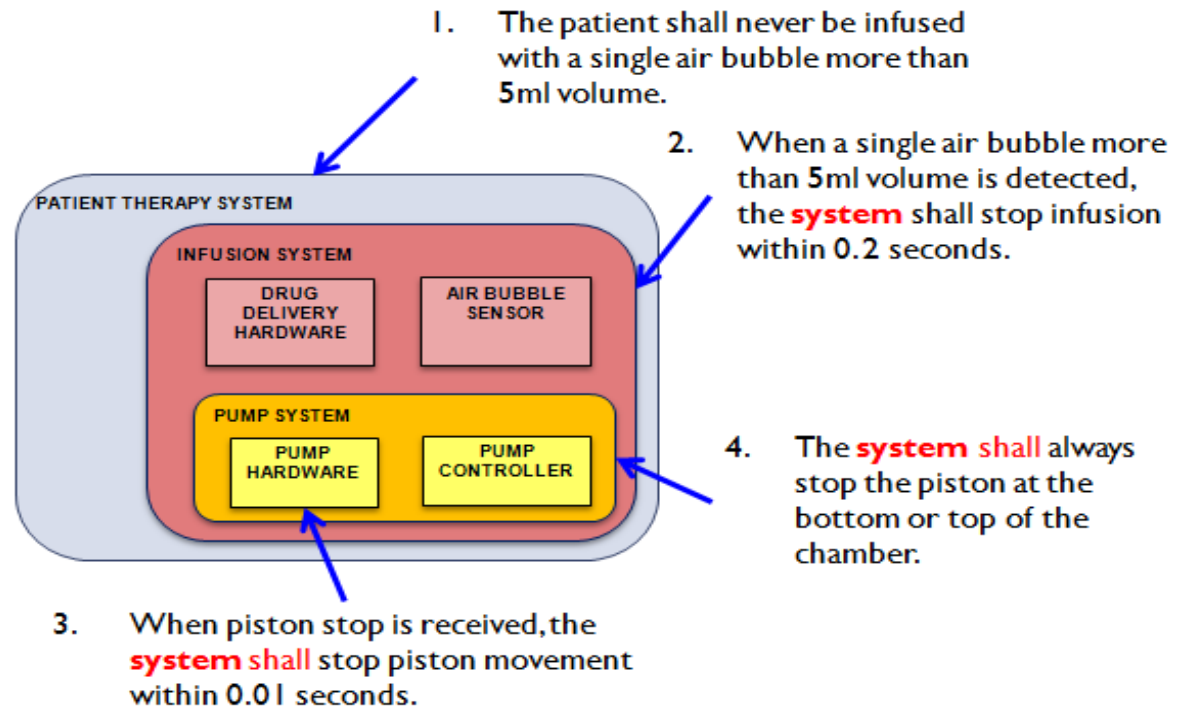
The patient shall never be infused with a single air bubble more than 5ml volume.

When a single air bubble more than 5ml volume is detected, the **system** shall stop infusion within 0.2 seconds.

When piston stop is received, the **system** shall stop piston movement within 0.01 seconds.

The **system** shall always stop the piston at the bottom or top of the chamber.

## Requirements and Design Information



Typical requirement documents span multiple levels of a system architecture

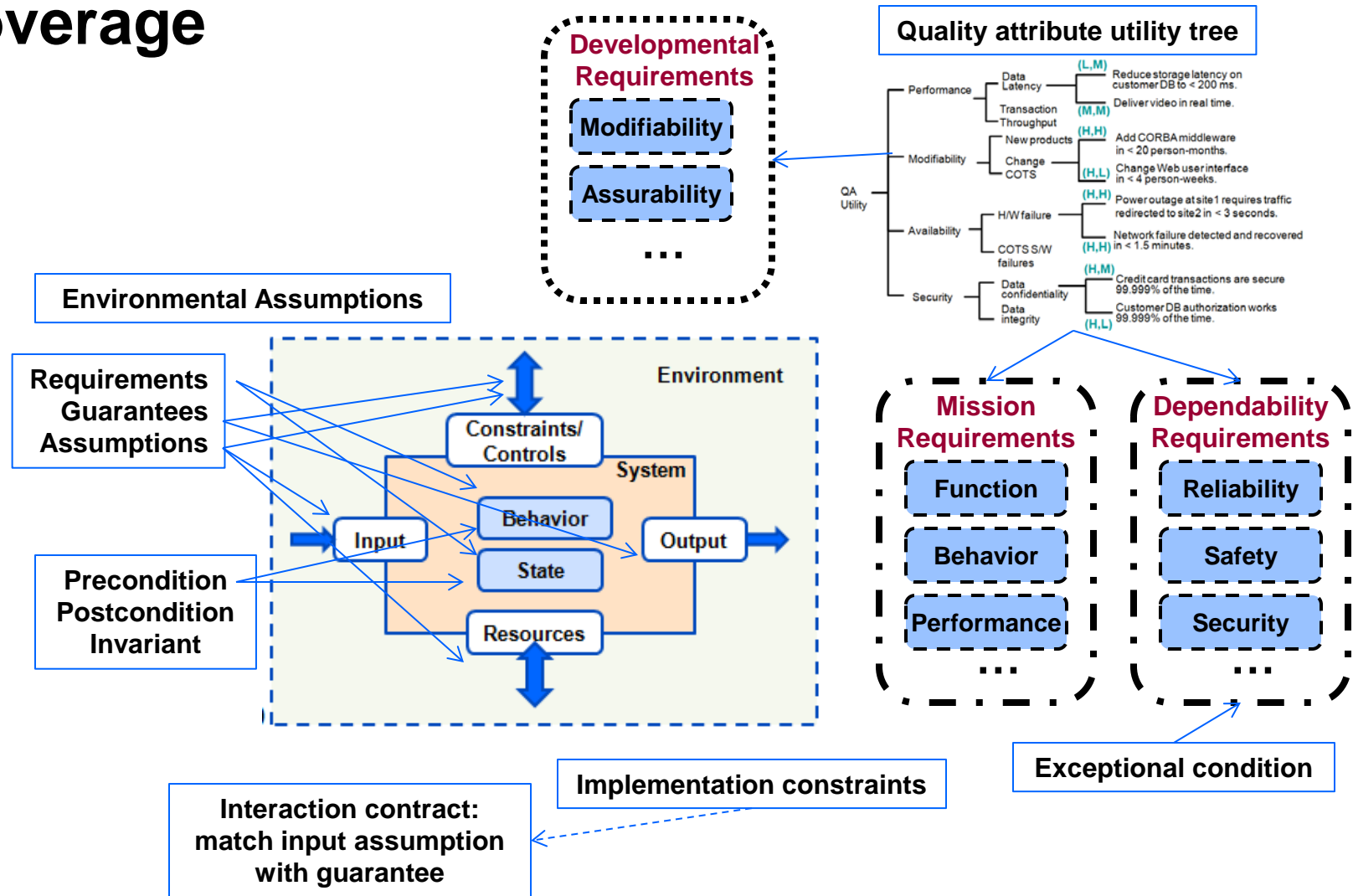
We have made architecture design decisions.

We have effectively *specified a partial architecture*

Adapted from M. Whalen presentation

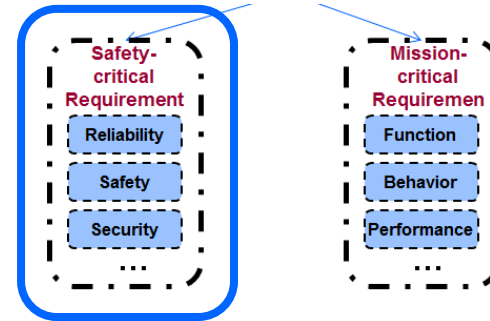
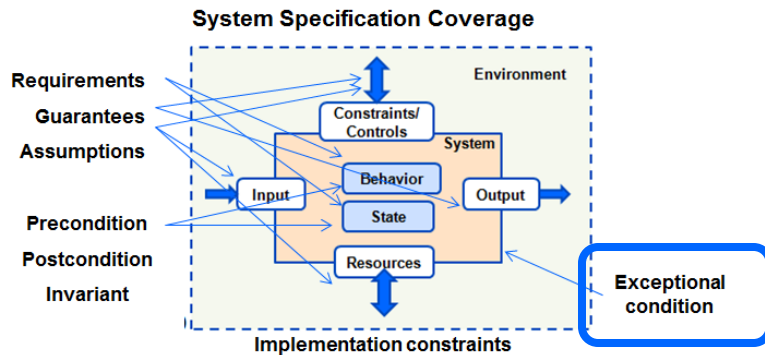


# System Specification and Requirements Coverage

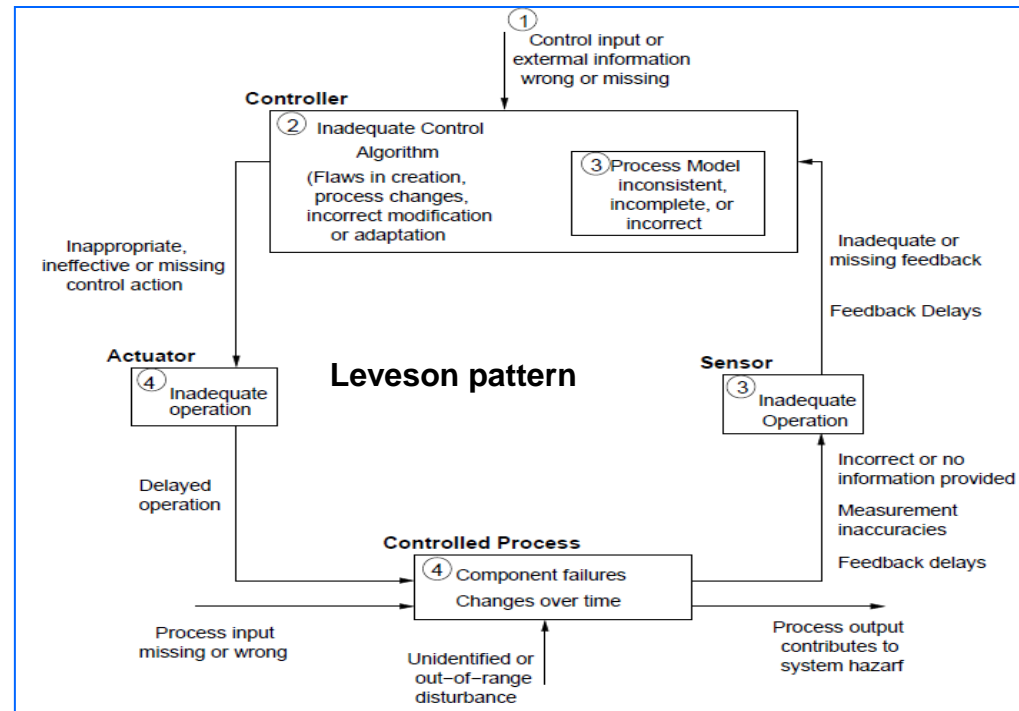
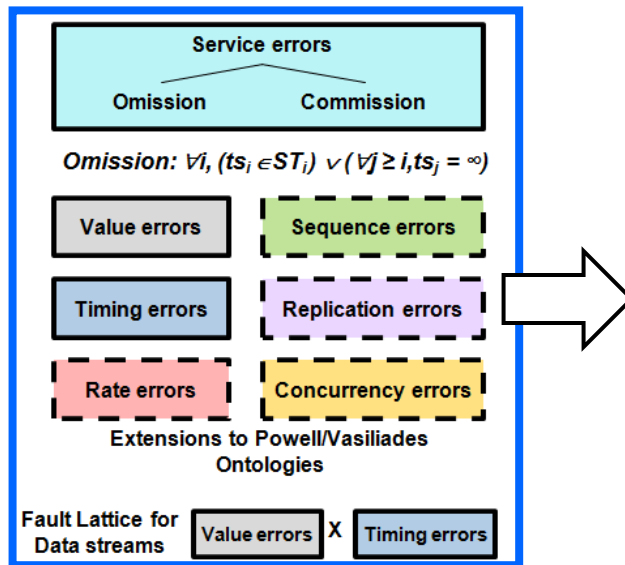




# Architecture-led Requirement & Hazard Specification



## Error Propagation Ontology



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Challenges in Safety-critical Software-intensive systems

An Architecture-centric Virtual Integration Strategy with SAE AADL

Improving the Quality of Requirements

▶ Architecture Fault Modeling and Safety

Incremental Life-cycle Assurance of Systems

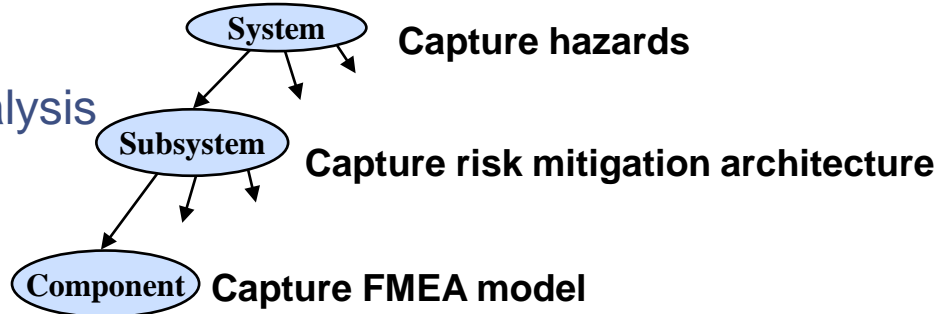
Summary and Conclusion



# AADL Error Model Scope and Purpose

System safety process uses many individual methods and analyses, e.g.

- hazard analysis
- failure modes and effects analysis
- fault trees
- Markov processes



Goal: a general facility for modeling fault/error/failure behaviors that can be used for several modeling and analysis activities.

Annotated architecture model permits checking for **consistency** and **completeness** between these various declarations.

Related analyses are also useful for other purposes, e.g.

- maintainability
- availability
- Integrity
- Security

*SAE ARP 4761 Guidelines and Methods for Conducting the Safety Assessment Process on Civil Airborne Systems and Equipment*  
***Demonstrated in SAVI Wheel Braking System Example***

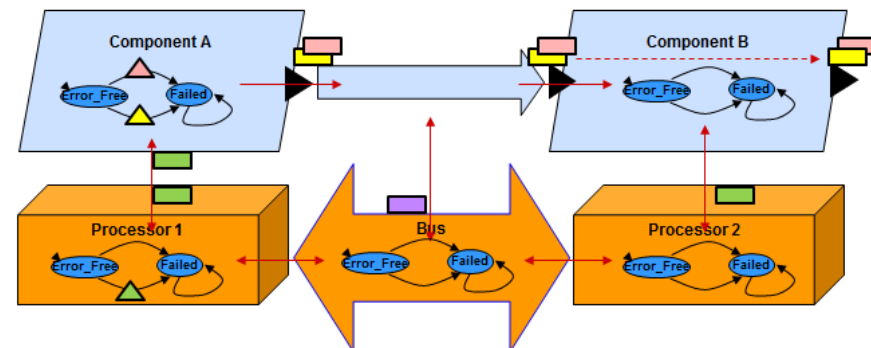
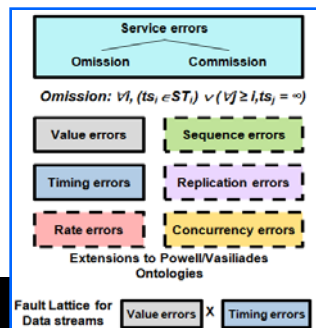
Error Model Annex can be adapted to other ADLs



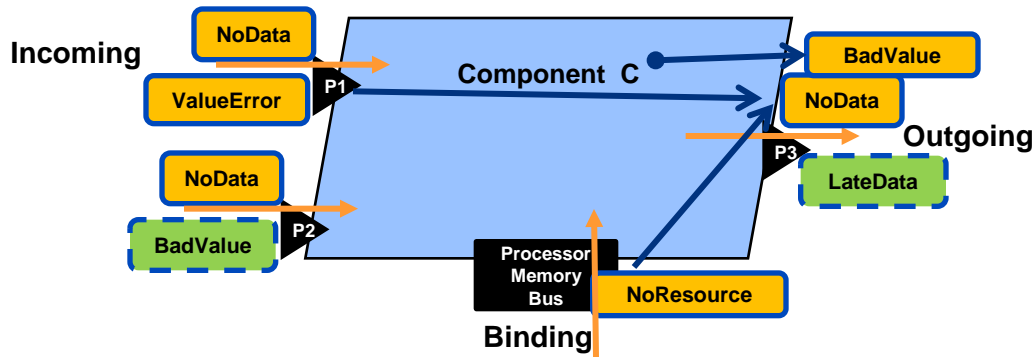
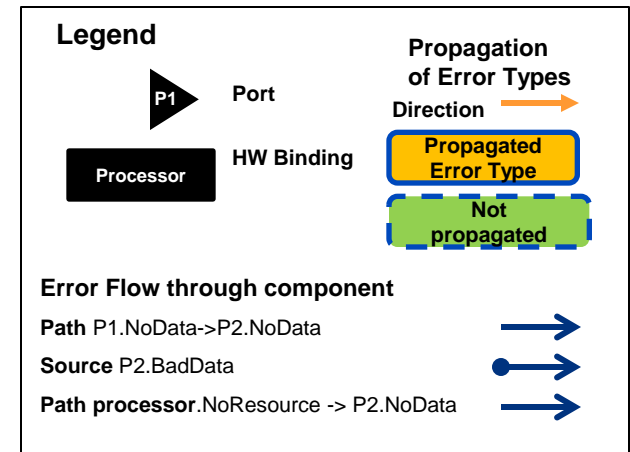
# Error Model V2: Abstraction and Refinement

Four levels of abstraction:

- Focus on fault interaction with other components
  - Probabilistic error sources, sinks, paths and transformations
  - Fault propagation and Transformation Calculus (FPTC) from York U.
- Focus on fault behavior of components
  - Probabilistic typed error events, error states, propagations
  - Voting logic, error detection, recovery, repair
- Focus on fault behavior in terms of subcomponent fault behaviors
  - Composite error behavior state logic maps states of parts into (abstracted) states of composite
- Types of malfunctions and propagations
  - Common fault ontology



# Error Propagation Contracts



“Not“ on propagated indicates that this error type is intended to be contained.  
 This allows us to determine whether propagation specification is complete.

## Incoming/Assumed

- Error Propagation  
Propagated errors
- Error Containment:  
Errors not propagated

## Outgoing/Contract

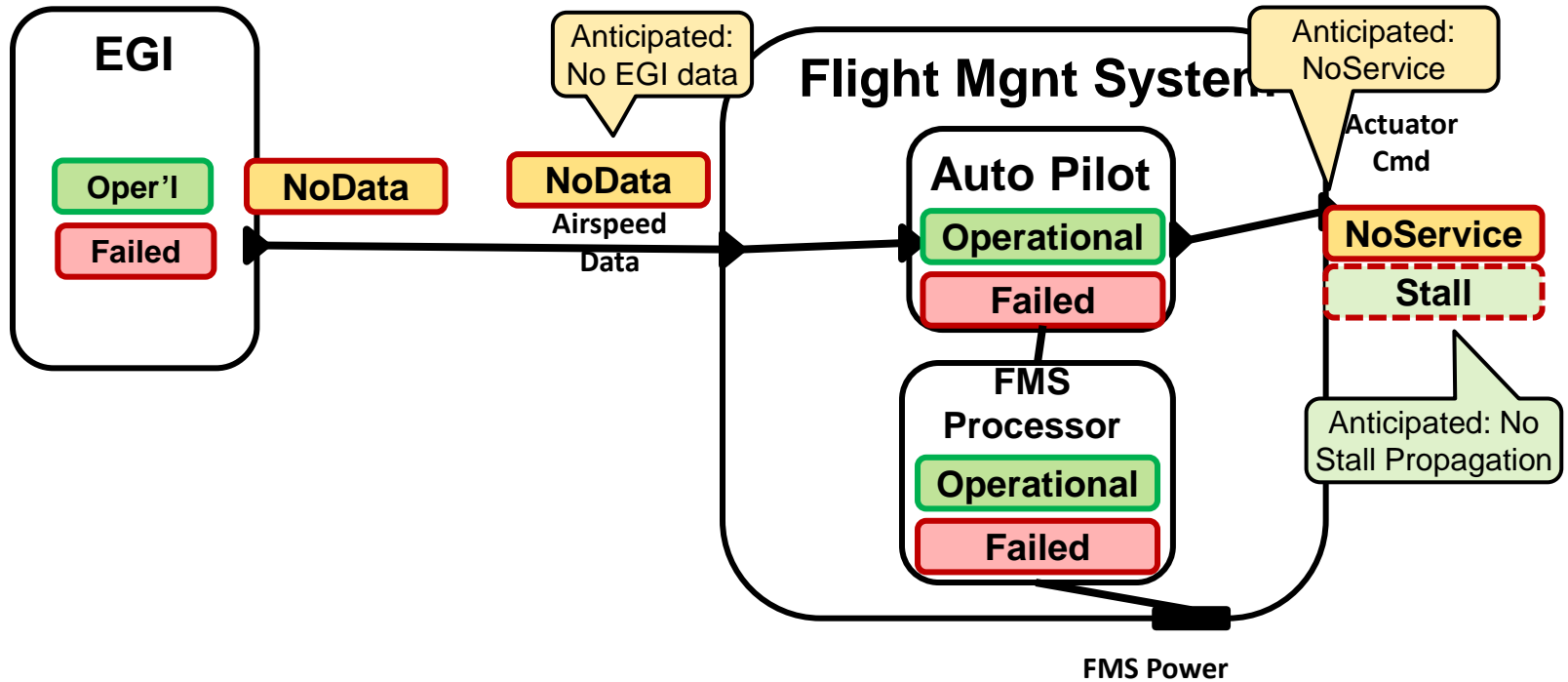
- Error Propagation
- Error Containment

## Bound resources

- Error Propagation
- Error Containment
- Propagation to resource



# Original Preliminary System Safety Analysis (PSSA)



System engineering activity with focus on failing components.

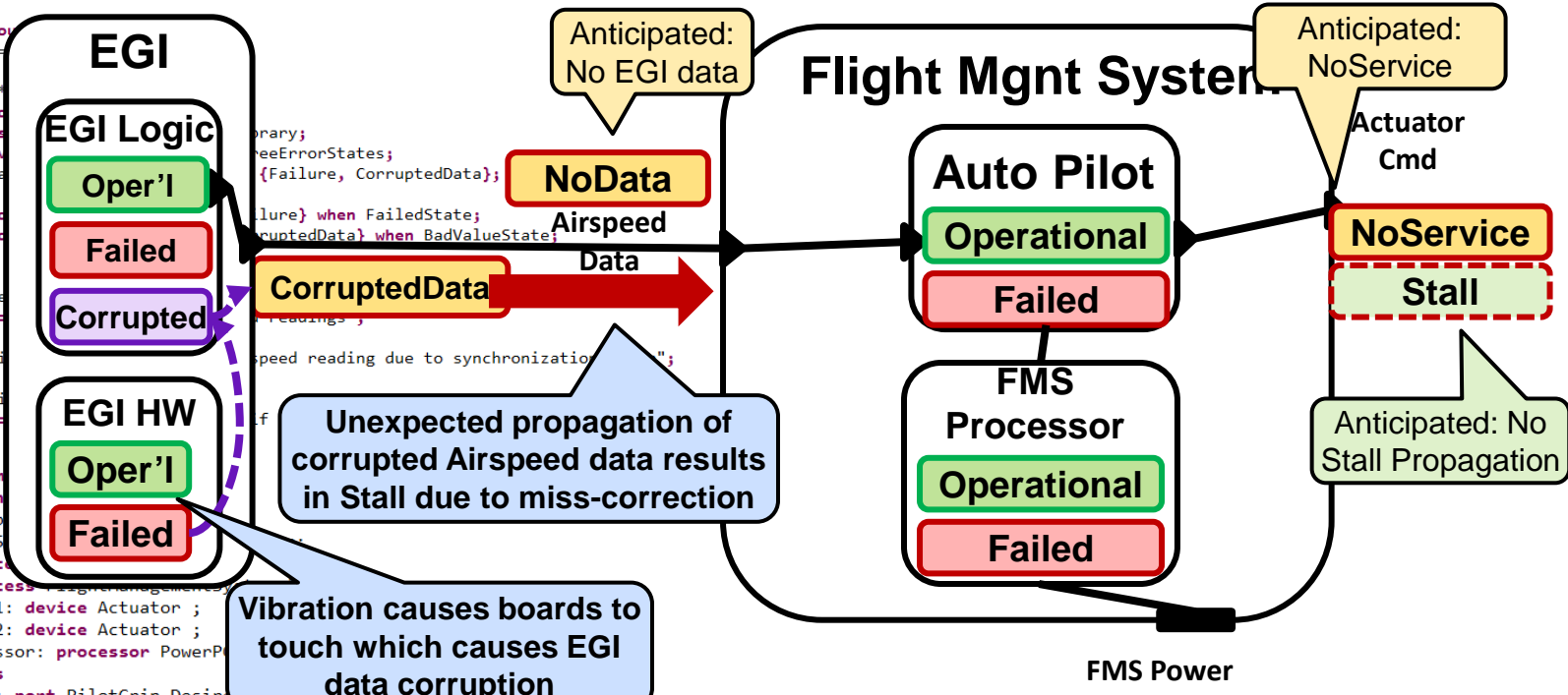


# Discovery of Unexpected PSSA Hazard through Repeated Virtual Integration

```

system EGI
features
  trueairspeed: out data port DataDictionary::Velocity;
flows
  f1: flow so
    Latency =
  };
annex EMV2 {
  error pro
  use types
  use behav
  truea
flows
  ef1:erro
  ef2:erro
properties
  EMV2::hazard
  [ crossref
  failure =
  phase =>
  descript
  severity
  critical
  comment =
system imple
subcomponen
  PilotGrip
  PositionS
  EGI: syst
  FMS: proces
  Actuator1: device Actuator ;
  Actuator2: device Actuator ;
  FMSProcessor: processor PowerP
connections
  pilotCmd: port PilotGrip.Desir
  sensedPosition: port PositionSensor.PositionReading -> FMS.Position;
  Actuator1Cmd: port FMS.ActCmd -> Actuator1.ActCmd;
  Actuator2Cmd: port FMS.ActCmd -> Actuator2.ActCmd;
  vtx: port EGI.TrueAirSpeed -> FMS.TrueAirSpeed;
f
  ⚠ Outgoing propagation (Failure, CorruptedData) is not handled. Expected incoming (Failure)
  / Actuator1Cmd -> Actuator1.FS
  {
    Latency => 15 ms .. 20 ms;
  };

```



EGI maintainer adds corrupted data hazard to model. Error Model analysis of integrated model detects unhandled propagation.



# Recent Automated FMEA Experience

Failure Modes and Effects Analyses are rigorous and comprehensive reliability and safety design evaluations

- Required by industry standards and Government policies
- When performed manually are usually done once due to cost and schedule
- If automated allows for
  - multiple iterations from conceptual to detailed design
  - Tradeoff studies and evaluation of alternatives
  - Early identification of potential problems

ID	Item	Initial State	Initial Failure Mode	1st Level Effect	Transition	2nd Level Effect	Transition	3rd Level Effect	Severity	M
1	Sat_Bus	Working	Failure	Failed		Failed	Recovery	Working		Workin
1	Sat_Payload	Working		Working	Bus failure causes payload transition	Standby		Standby	Bus Recovery Causes Payload Transition	Workin
2	Sat_Bus	Working		Working		Working	5			
2	Sat_Payload	Working	Failure	Failed	Recovery	Working	5			

Largest analysis of satellite to date consists of 26,000 failure modes

- Includes detailed model of satellite bus
- 20 states perform failure mode
- Longest failure mode sequences have 25 transitions (i.e., 25 effects)

Myron Hecht, Aerospace Corp.  
Safety Analysis for JPL, member of DO-178C committee





# Support of SAE ARP4761 System Safety Assessment Practice



**FHA**  
Spreadsheet  
Uses error sources

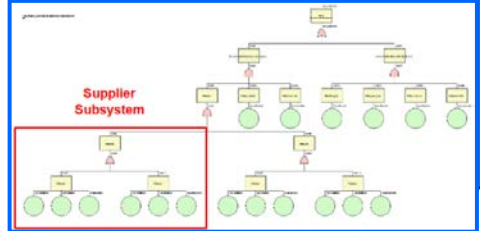
Component	Error	Hazard Description	Crossrefer	Functional Failure	Operational P
StabilatorPositionSel	"ServiceOmission of	"No stabilator position readings due to s	"1.1.3"	"Loss of sensor readings"	"all"
StabAct1	"ServiceOmission of	"Failure to move stabilator into desired	"1.1.2"	"Loss of actuator functionalit	"all"
StabAct2	"ServiceOmission of	"Failure to move stabilator into desired	"1.1.2"	"Loss of actuator functionalit	"all"
StabilatorController	"null on ActCmd"	"Absence of computed data should signa	"1.1.1"	"Loss of guidance values"	"Approach"
StabilatorController	"null on ActCmd"	"Absence of computed data should signa	"1.1.1"	"Loss of guidance values"	"Approach"
StabilatorController	"null on ActCmd"	"Absence of computed data should signa	"1.1.1"	"Loss of guidance values"	"Approach"

**Markov Chain**  
PRISM  
Uses error flows & behavior

**FMEA**  
Spreadsheet  
Uses error flows & propagations

Three CPU FMEA						
Item	Initial State	Initial Failure	Lat Level Effect	Transition	Final Level Effect	
CPU_1.cpu	ErrorFree	CPU_failure	PermanentError	out_CPU_Failed	PermanentError	
PR_AP_L	ErrorFree		ErrorFree	CPU_Failed(N)	PermanentError	
CPU_2.cpu	ErrorFree		ErrorFree		ErrorFree	
PR_AP_R	ErrorFree		ErrorFree		ErrorFree	
PR_FGS_L1	ErrorFree		ErrorFree		ErrorFree	
CPU_3.cpu	ErrorFree		ErrorFree		ErrorFree	
PR_FGS_R1	ErrorFree		ErrorFree		ErrorFree	
CPU_1.cpu	ErrorFree		ErrorFree		ErrorFree	
PR_AP_L	ErrorFree		ErrorFree		ErrorFree	
CPU_2.cpu	ErrorFree	CPU_failure	PermanentError	out_CPU_Failed	PermanentError	
PR_AP_R	ErrorFree		ErrorFree	CPU_Failed(N)	PermanentError	
PR_FGS_L1	ErrorFree		ErrorFree	CPU_Failed(N)	PermanentError	
CPU_3.cpu	ErrorFree		ErrorFree		ErrorFree	
PR_FGS_R1	ErrorFree		ErrorFree		ErrorFree	

**FTA**  
CAFTA, OpenFTA  
Uses composite error behavior



**RBD/DD**  
OSATE plugin  
Uses composite error behavior



# The Symptom: Missed Stepper Motor Steps

Stepper motor (SM) controls a valve

- Commanded to achieve a specified valve position
  - Fixed position range mapped into units of SM steps
- New target positions can arrive at any time
  - SM immediately responds to the new desired position

Safety hazard due to software design

- Execution time variation results in missed steps
- Leads to misaligned stepper motor position and control system states
- Sensor feedback not granular enough to detect individual step misses

## Software modeled and verified in SCADE

Full reliance on SCADE of SM & all functionality  
Problems with missing steps not detected

## Software tests did not discover the issue

Time sensitive systems are hard to test for.

## Two Customer Proposed Solutions

Sending of data at 12ms offset from dispatch  
Buffering of command by SM interface  
No analytical confidence that the problem will be addressed

## Other Challenge Problems

Aircraft wheel braking system  
Engine control power up  
Situational Awareness & health monitoring



# Outline

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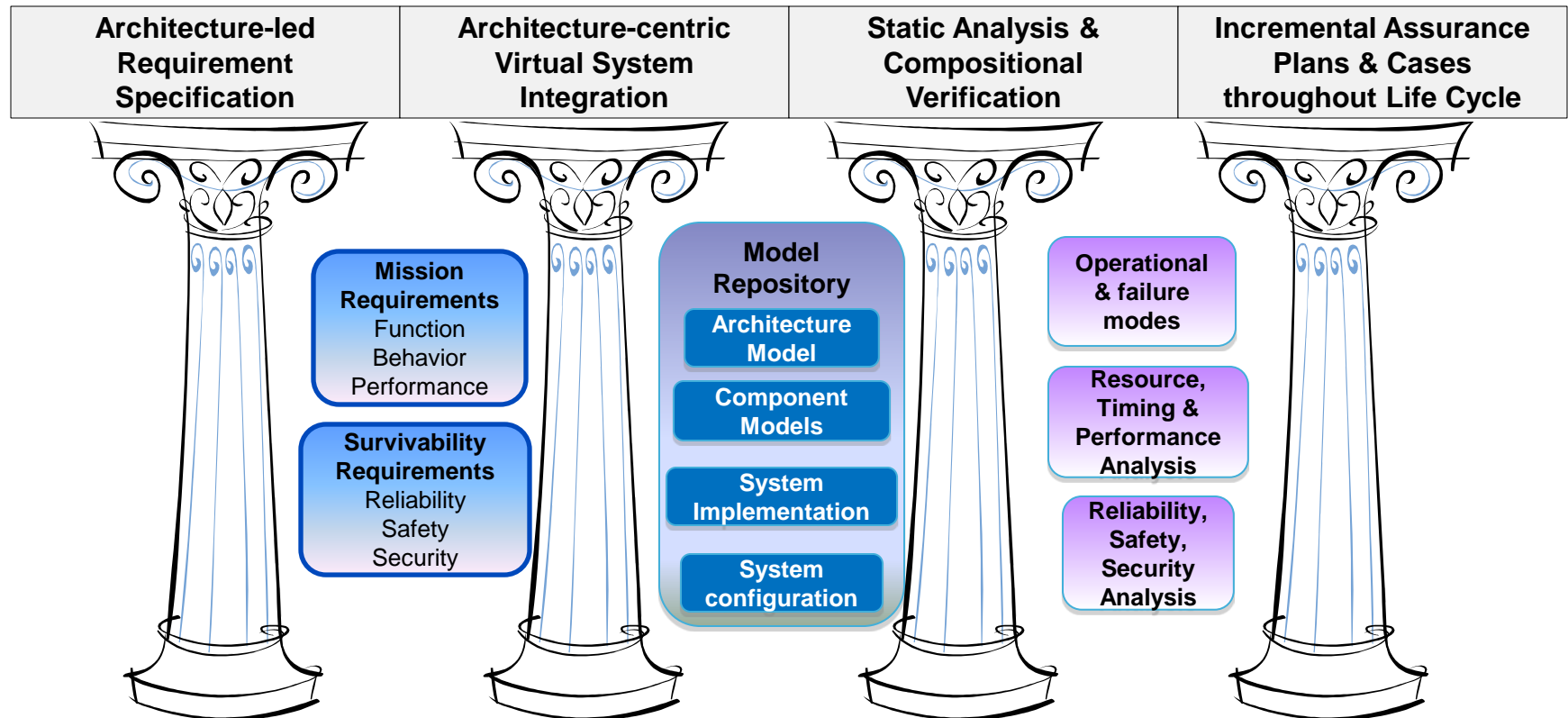
▶ Incremental Life-cycle Assurance of Systems

Summary and Conclusion



# Reliability & Qualification Improvement Strategy

2010 SEI Study for AMRDEC  
Aviation Engineering Directorate



**Four pillars for Improving Quality of Critical Software-reliant Systems**



# Contract-based Compositional Verification

## Secure Mathematically-Assured Composition of Control Models

### Key Problem

Many vulnerabilities occur at component interfaces.  
How can we use formal methods to detect these vulnerabilities and build provably secure systems?

TA4 – Research Integration and  
Formal Methods Workbench  
Rockwell Collins and  
University of Minnesota

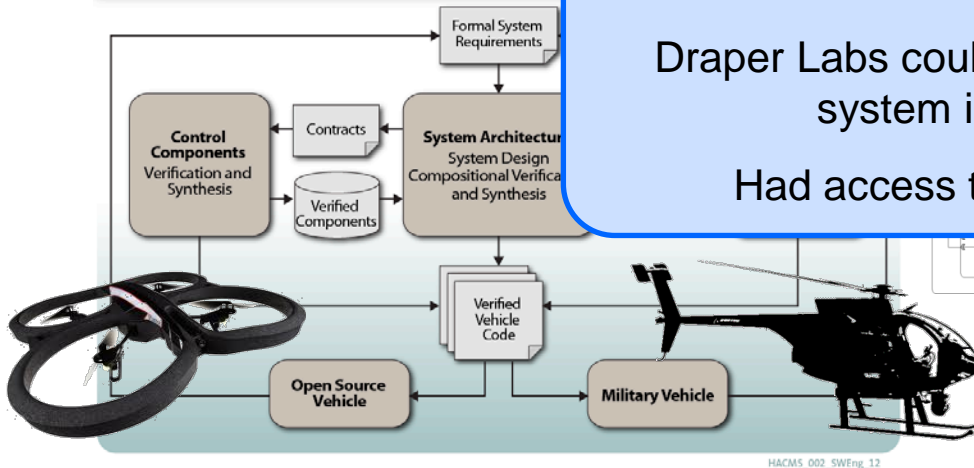


16 months into the project

Draper Labs could not hack into the  
system in 6 weeks

Had access to source code

### ARCHITECTURE-CENTRIC PROOF



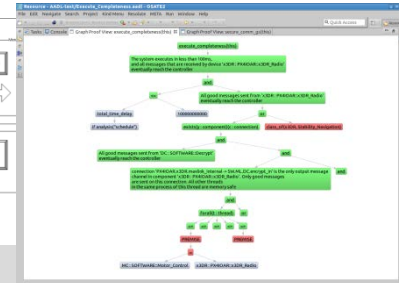
HACMS\_002\_SWEng\_12

### Technical Approach

- Develop a complete, formal architecture model for UAVs that provides robustness against cyber attack
- Develop compositional verification tools driven from the architecture model for combining formal evidence from multiple sources, components, and subsystems
- Develop synthesis tools to generate flight software for UAVs directly from the architecture model, verified components, and verified operation system

### Accomplishments

- Created AADL model of vehicle hardware & software architecture
- Identified system-level requirements to be verified based on input from Red Team evaluations
- Developed Resolute analysis tool for capturing and evaluating assurance case arguments linked to AADL model
- Developed example assurance cases for two security requirements
- Developed synthesis tool for auto-generation of configuration data and glue code for OS and platform hardware



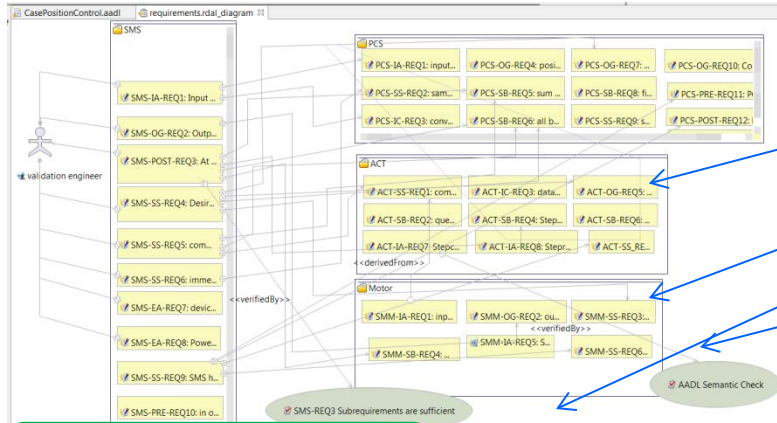
Software Engineering Institute

Carnegie Mellon

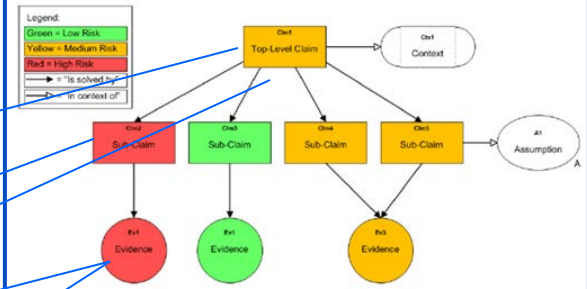
Feiler, C  
© 2014 Carnegie Mellon

Open source tools available at  
[github.com/smaccm](https://github.com/smaccm)

# Integrated Approach to Requirement V&V through Assurance Automation



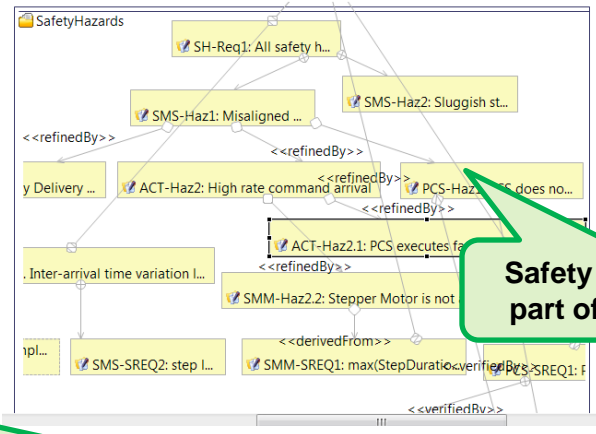
**Requirement coverage  
Assumption evidence**



**Generated  
assurance cases**

Element	Verified	Level (%)	Risk
Requirements Group SafetyHazards		NaN	
Requirements Group ACT		NaN	
Requirement ACT-SB-REQ2: queue size zero and abort overflow		100.0	Low
Requirement ACT-SS-REQ9: Homing command results in SMM		NaN	High
Requirement ACT-OG-REQ5: MaxStepCount of 15 is used as step		100.0	Low
Requirement ACT-SB-REQ6: StepCount == zero when reset to n		100.0	Low
Requirement ACT-IA-REQ7: Stepcount within range		NaN	High
Requirement ACT-SS-REQ1: command arrival driven command		100.0	Low
Requirement ACT-SB-REQ4: StepCount == # of step signals to n		NaN	High
Requirement ACT-IA-REQ8: StepCount == M		NaN	High
Requirement ACT-IA-REQ9: StepCount == M		100.0	Low
Requirement ACT-IA-REQ10: StepCount == M		NaN	High

**Evidence records in terms of claims that requirements have been met**



**Safety hazards are part of the picture**

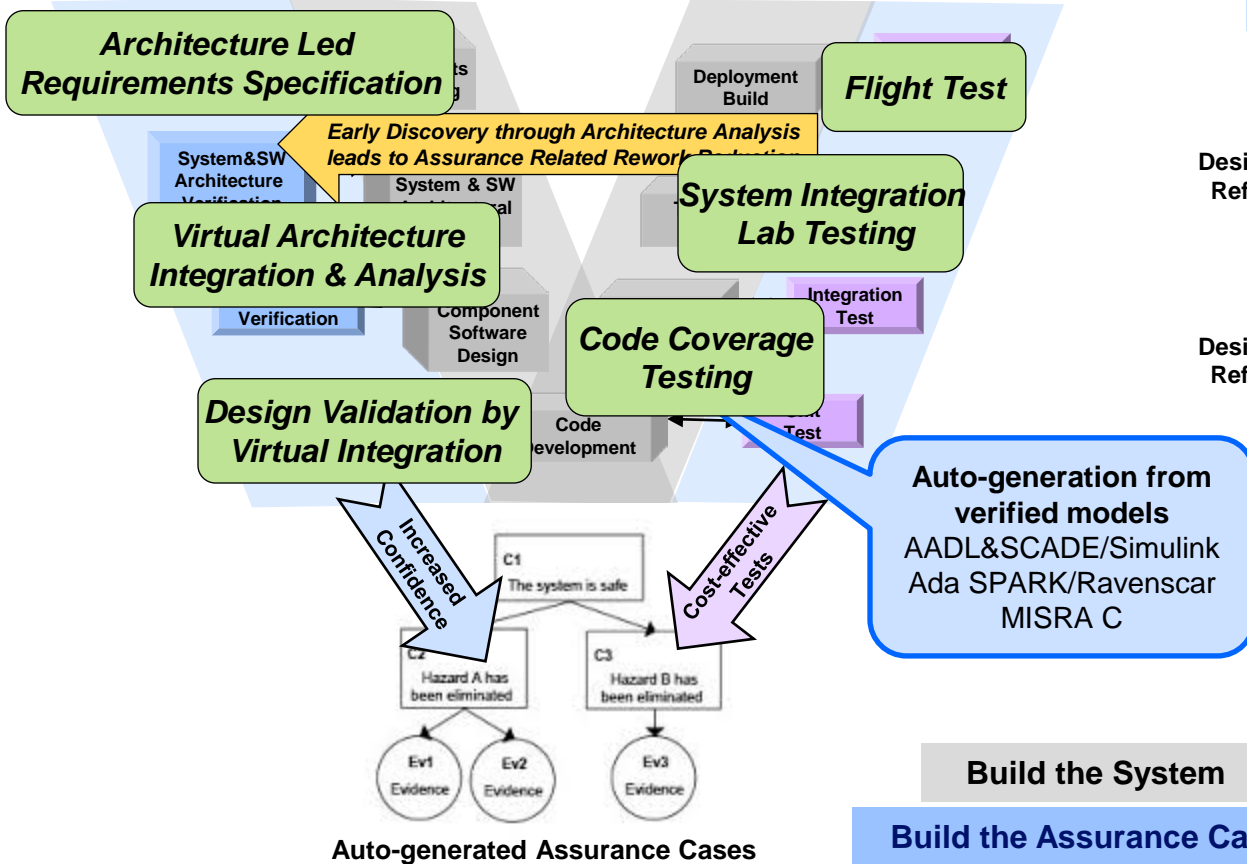
**Linkage to automated test harnesses**



# Building the Assurance Case throughout the Life Cycle

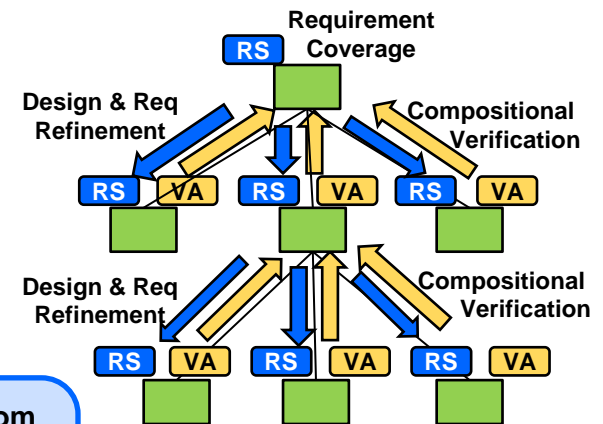
Continuous Confidence Measure throughout Life Cycle that a System Meets its Requirements

Architecture-centric Virtual Integration



Incremental Evolution and Execution of Assurance Plans

Incremental Architecture & Requirement Evolution



Incremental Contract-based Compositional Verification

Build the System

Build the Assurance Case



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# Benefits of Architecture-centric Engineering

## Reduce risks

- Analyze system early and throughout life cycle
- Understand system wide impact
- Validate assumptions across system

## Increase confidence

- Validate models to complement integration testing
- Validate model assumptions in operational system
- Evolve system models in increasing fidelity

## Reduce cost

- Fewer system integration problems
- Fewer validation steps through use of validated generators



# References

AADL Website [www.aadl.info](http://www.aadl.info) and AADL Wiki [www.aadl.info/wiki](http://www.aadl.info/wiki)

Blog entries and podcasts on AADL at [www.sei.cmu.edu](http://www.sei.cmu.edu)

AADL Book in SEI Series of Addison-Wesley

<http://www.informit.com/store/product.aspx?isbn=0321888944>

On AADL and Model-based Engineering

[http://www.sei.cmu.edu/library/assets/ResearchandTechnology\\_AADLandMBE.pdf](http://www.sei.cmu.edu/library/assets/ResearchandTechnology_AADLandMBE.pdf)

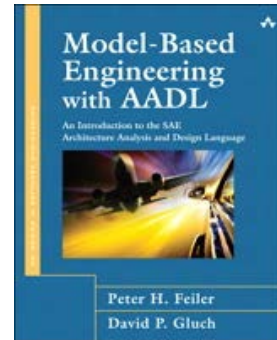
On an architecture-centric virtual integration practice and SAVI

[http://www.sei.cmu.edu/architecture/research/model-based-engineering/virtual\\_system\\_integration.cfm](http://www.sei.cmu.edu/architecture/research/model-based-engineering/virtual_system_integration.cfm)

On an a four pillar improvement strategy for software system verification and qualification

<http://blog.sei.cmu.edu/post.cfm/improving-safety-critical-systems-with-a-reliability-validation-improvement-framework>

Webinars on system verification <https://www.csiac.org/event/architecture-centric-virtual-integration-strategy-safety-critical-system-verification> and on architecture trade studies with AADL <https://www.webcaster4.com/Webcast/Page/139/5357>





15 Years of the SAE AS-2C AADL Committee

10 Years since the first publication of the SAE AADL standard

And many more 😊



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