NSRP No. 0297 March 1990

### THE NATIONAL SHIPBUILDING RESEARCH PROGRAM

FLAME BENDING OF PIPE FOR ALIGNMENT CONTROL

PANEL SP-7 PROJECT REPORT

This project was performed by Puget Sound Naval Shipyard under sub-contract from Ingalls Shipbuilding, Inc. Funds were provided by the U. S. Navy and the Maritime Administration of the U. S.

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#### FOREWORD

This project presents the results of an R&D project initiated by members of the Society of Naval Architects and Marine Engineers Ship Production Committee, Panel SP-7.

The project was conducted by Puget Sound Naval Shipyard by Steve Nelson, J. Dwight, Dale Heagy, D. Mortvedt, Bob Houghteling, Frank Gatto (SP-7 member) and D. Coglizer. SP-7 Panel Chairman was Lee Kvidahl and Program Manager was O. J. Davis, both of Ingalls Shipbuilding, Inc. The NSRP Administrator was Virgil Rinehart, Maritime Administration of the Department of Transportation. A Word About the NSRP:

The National Shipbuilding Research Program (NSRP) has been engaged in research related to improvements in shipbuilding in the U. S. since 1973. The program is a cooperative effort involving commercial and U. S. Naval shipyards and related agencies, industries and educational institutions.

Since the inception of the program in 1973 R&D projects have been performed with significant contribution in the areas of facilities, environmental effects, outfitting and production aids, design and production integration, human resource innovatior shipbuilding standards, welding, industrial engineering, educatior and training, flexible automation and surface preparation and coatings. A library and bibliography of NSRP reports is maintained at the University of Michigan, Transportation Research Institute, Ann Arbor, Michigan.

The program is funded by cooperative agreement contracts by the U. S. Navy and the Maritime administration of the U. S. Department of Transportation.

#### ABSTRACT

The principles of flame straightening, long in use on plate structures in shipbuilding, have been applied to the problem of precision alignment of fluid system piping in shipbuilding and overhaul. Reduction of residual stresses by elimination of mechanically applied stresses to pipes for alignment prior to welding or bolting in place is a desirable objective.

This project is a first effort to develop techniques of heat control and patterns of heating to achieve alignment without adverse effects to base metal. Extensive testing has been performed and results documented to provide a data base for refir ment of procedures to be used in ship production and overhaul. It was not possible within available time and funding to reach definitive conclusions on CRES pipes, however for carbon steel and copper-nickel alloys, the report shows positive results. No significant detrimental effects of repeated controlled heati: were found and sufficient bending is produced to warrant use of flame bending of carbon steel and copper-nickel pipe in shipbuilding.

> O. J. Davis SP-7 Program Manager

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#### APPENDICES

Appendix "A"CODAS Calibration Data Gathering Method - Keystroke Commands.A-1 to A-2Appendix "B"Physical Tests and Metallography of Flame Bent PipeB-1 to B-8Appendix "C"Time vs Deflection Charts for Flame Bending of PipeC-1 to C-9

#### 1 BACKGROUND

#### 1.1 Overview

This report describes the detail laboratory, analytical and hands-on work performed. The effects of flame bending on pipe movement are examined in detail. The effects of flame bending on metallurgical and mechanical properties are also reported.

All empirical data (derived by experimentation) has been compiled and is located in the appendix. Simple pipe movements can be calculated. A method of torch heat calibration has been developed using precision flow meters.

This detail study should form the basis of a practical shop document to be written, which allows the mechanic to perform flame bending with some assurance of a predictable result without damage to the pipe system material.

The work has been funded by the SP-7 Welding Panel.

It is anticipated that a brief practical Welding Journal Article may be written and submitted for publication during 1990.

#### 1.1.1 Typical Flame Bent Pipe

For those panel members not familiar with flame bending pipe the following sketches show some of the fundamentals.



#### PIPE LAYED OUT FOR VEE HEAT BENDING

The bending heat pattern shown below requires two torches be used simultaneously. The heating pattern starts at the VEE TIP HINGE and proceeds toward the wide section of the VEE PATTERN.



PIPE CUT AND SPREAD OUT SHOWING DUAL VEE HEAT PATTERN

#### 1.1.1.1 Final Vee Heat Configuration

The final result is a pipe bent around a HINGE point. Figure 1.1.1.1 below can be generated using the HOLT or Line Heat method.

AFTER HEATING WITH VEE TORCH PATTERN



FILE:VEE\_HT\_3

FIGURE 1.1.1.1

#### 1.2 Literature Review

A detailed literature survey has been completed. The survey was performed through the Edison Welding Institute by the Battelle Institute, Columbus Ohio. Dr. H. W. Mishler, Senior Research Engineer provided abstracts of 95 citations, most of which were in German. Complete copies of four (4) Ship Structure Committee reports were provided from. Dr. Mishlerts personal library. These reports deal in detail with the effects of flame straightening on ships structures.

#### 1.2.1 Primary Concepts, Richard E HOLT

Professor Richard E. Holt has written the definitive article on flame bending of structural shapes. This article explains the principals of thermal stress and strain along a single axis (perfect confinement), along a single axis (variable confinement) and biaxial restraint.

Professor Holt identifies the spot heat as the basis of all heat patterns. "Consider a circular spot heated to an elevated temperature. The radial and tangential stresses in the hot zone will be compression equal to 0.5  $\alpha \ \Delta TE$ . In the restraining material the radial stress will be compression with the tangential-stress tension as shown below:

radial 
$$\sigma_r = -0.5 \propto \Delta TE \frac{a^2}{r^2}$$
  
tangential  $\sigma_l = -0.5 \propto \Delta TE \frac{a^2}{r^2}$ 

where E = modulus of elasticity; a = radius of heated zone; r = radius at which stress is calculated; AT = temperature difference between hot zone and restraining metal; a= coefficient of thermal expansion." 2

The concepts of spot heats, line heats (traveling spot) and vee heats are examined in detail as related to structural shapes such as angle, channel and I-beams.

Holt has determined that "the basic heat pattern for producing a bend in metal objects is the vee heat". 3 professor Richard E. Holt quantifies this original work by Joseph Holt into a set of general rules and simple equations.

1 Holt, R. E., "PRIMARY CONCEPTS FOR FLAME BENDING", Welding Journal, June 1971. 2 Holt, R. E., "PRIMARY CONCEPTS FOR FLAME BENDING", Welding Journal, June 1971. 3 Holt, Joseph, Contraction as a Friend in Need, Joseph Holt, 1938. Mechanical properties of structural steels were shown to improve after flame bending. Yield strengths were slightly above the original values and Ultimate tensile strengths showed slight reductions. "Properly placed heat patterns produce a residual stress pattern that is lower in magnitude and more uniform than as-rolled material. Tests conducted by the US Army Corps of Engineers at Clear Alaska, showed the flame straightened A36 beams were superior to the as rolled beams."

#### 1.2.2 Flame Straightening & its Effect on Base Metal, SSC-198 <sup>5</sup>

This report is essentially a summary of literature and basic metallurgy associated with flame straightening on various classes of ship building plate steels. This document lists 70 references.

### 1.2.3 Effect of Flame & Mechanical Straightening on Properties of Weldments, SSC-207

This research covered various steel plate in the range from 40,000 psi tensile to 100,000 psi tensile strength. One of the findings was;

\* "Buckling-type distortion decreases drastically as the plat thickness increases, and it almost disappears when the plate thickness exceeds about 3/8"."

Fracture transition temperatures were determined for ABS-2 type plat in the as received and in the flame straightened condition. Flame straightening temperatures were very close to the experimental temperatures of the PUGET SOUND NAVAL SHIPYARD work. Their temperature were 1300 to 1400 degrees F.

As received ABS-2 plate have fracture transition temperatures of +52 F. Flame straightened plate in this study had fracture transition temperatures of +70 F.

Their conclusion was that "widespread use of flame straightening on ABS-2 steel is an acceptable procedure". The carbon steel pipe used in the Naval Shipyards are very similar to ABS-2 grade steel.

<sup>4</sup> Holt, R. E., "PRIMARY CONCEPTS FOR FLAME BENDING", Welding Journal, June 1971.

<sup>5</sup> H. E. Pattee, R. M. Evans, and R. E. Monroe., "Flame Straightening and its Effect on Base Metal Properties", Ship Struct Committee Report No. 198 (August 1969)

<sup>6</sup> H. E. Pattee, R. M. Evans, and R. E. Monroe., "Effect of Flame & Mechanical Straightening on Material Properties of Weldments", Ship Structure Committee Report No. 207 (1970)

#### **2 OBJECTIVES**

#### 2.1 Shop or Field Calibration Method

To Develop an accurate and simple method of calibrating the heat output of the flame bending heat source.

#### 2.2 Travel Speed and Temperature

To determine the relationship between torch travel speed and temperature gradient

#### 2.3 Vee Heat Pattern Versus Pipe Deflection

To determine the relationship between Vee heat pattern and pipe deflection.

#### 2.4 Multiple Heat Effects

To determine the effectiveness of multiple heats.

#### 2.5 Experimental Process Measurements

To determine a practical method of monitoring the heat distribution, using critical process parameters such as, torch travel speed, path, spot diameter and torch heat output.

#### 2.6 Practical Pipe Temperature Measurement

To investigate practical methods for measuring temperatures in process on the flame side, below red hot, for field applications.

#### 2.7 Metallurgical & Mechanical Property Effects .

To determine any detrimental effects of flame bending on the mechanical properties and metallurgical structure of various alloys.

#### 2.8 Internal Sea Water Contamination Effects

To determine any detrimental effects of flame bending on pipe with service related internal contamination.

#### 2.9 Final Engineering Report

To write a detailed final engineering report.

#### 2.10 Shop Working Instruction

To develop a practical shop working instruction for use on the production shop floor.

#### 3 MATERIAL, EQUIPMENT & LABOR REQUIREMENTS

The primary purpose of this section is to describe in detail the experimental setup, material, equipment and recording devices used to achieve the final results.

#### 3.1 Material

#### 3.1.1 Material Type Tested .

The alloys tested were:

Alloy	Number of test heats
A. Carbon steel	57
B. Copper Nickel (70/30)	72
C. Copper Nickel (90/10)	76
D. Stainless Steel	15
E. Monel	5

TOTAL

225

#### 3.1.2 Pipe Material Specifications (new pipe)

All pipe was purchased new from the stock system with the exception of the seawater contaminated pipe described in section 3.1.3. Pipe diameter ranged from 2" to 12", with wall thickness from 1/8" to 3/4".

The use of stock system pipe material should represent a data base line which can be translated for use by other Naval Shipyards.

#### 3.1.3 Seawater Service Pipe Removal (12" & 4.5" Diameter)

Actual submarine main sea water (12" diameter) and auxiliary sea water (4.5" diameter) pipe represents a worst case condition for contaminated pipe.

The 4.5 and 12 inch seawater pipe was subsequently removed in lengths comparable to the new pipe lengths.

The pipe ends were sealed to preserve the internal contaminants until the actual test work was performed.

#### 3.2 Labor

#### 3.2.1 Code 138 Welding Engineers

All hands on lab work was performed by J. Dwight, Mr. Dale Heagy, Mr. Steve Nelson and Mr. Bob Houghteling. Mr. Derek Mortvedt assisted during certain phases of the work and was especially helpful on final review of results.

Mr. Frank Gatto was the Project Manager and provided detail technical guidance. Mr Steve Nelson and J. Dwight shared Project Engineer responsibilities. Mr. Douglas Coglizer provided financial and technical guidance for the entire program.

#### 3.2.2 Shop 06 Electronics

Mr. Tony D' Andrea of Shop 67 performed all electronic support work. Mr. D'Andrea performed all installation and setup of analog to digital conversion boards and performed the initial calibrations of LVDT 's.

#### 3.2.3 Code 260.2 Design Engineering Support

Code 260.2, Design Engineering, wrote a detail instruction to have 4.5 and 12 inch diameter Main Seawater Copper Nickel (70-30) pipe removed from a submarine. They selected EX-SSN618 which has been service for approximately 20 years.

Code 260.2 personnel have been helpful and maintained an interest i the flame bending study throughout the program.

#### 3.3 Equipment

3.3.1 Heat Source



#### 3.3.1.1 Oxy-Fuel Torch

A standard MAPP GAS + Oxygen Fuel torch was used through all tests. This gas is the standard used in the Shipyard.

#### 3.3.1.2 OXY-Fuel Flow Meters

Flow rates are set, monitored and controlled by precision flow meters with "Y" fittings on the output side of the flow meter.

This configuration assures that each torch receives the same volume of gas.

#### 3.3.2 Software

#### 3.3.2.1 Lotus Freelance

Lotus "Freelance" was used to place all drawings directly into the word processing program.

#### 3.3.2.2 CODAS Program SoftWare

Analog to digital data conversions of pipe motion was recorded in 3 axis. The data was manipulated using the CODAS software. The results of these real time records were fed into lotus 1,2,3 for graph generation.

The finish graphs are contained in -- appendix C.

#### 3.3.3 .Electronic

#### 3.3.3.3 Real-time Waveform Scroller Board

The CODAS WFS-200 card was used to display the data in real time .as it was gathered. This card works with the METRA BYTE EXP-16 A-D conversion card (see section 3.3.3.2)

Pipe movement was recorded as each experiment proceeded. The data was real time on the computer screen with up to 8 windows visible at one time.

The CODAS real-time waveform scroller has software that allows the user to display data at different rates. The user can also look at very small segments of the data if required.

The CODAS circuit card and software eliminates the requirement for the user to write programs. Programming is required if the METRA BYTE EXP-16 is used with out support hard/software.

#### 3.3.3.2 Analog to Digital Conversion Card for the Compaq Computer

The METRA BYTE EXP-16 card was installed and used in conjunction with the real time waveform scroller board.

The function of this board is to convert the low voltage DC signals from various transducers such as LVDT'S and thermocouples to digital signals which the computer can process.

#### 3.3.3.3 Linear Voltage Displacement Transducers (LVDT)

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The LVDT's used in all tests for measuring deflection of pipe wer DC powered. These devices use a +/- 15VDC power supply and produc a DC output of +/- 10V. This output is subsequently reduced to +/- 5VDC for use with the CODAS BOARD.



PLUNGER MOVES IN BOTH DIRECTIONS

A general layout of the overall experimental gathering method can be seen in the below sketch.



#### 3.4 Mechanical Test Setup

The overall mechanical test setup is shown below.



#### 3.4.1 Thermocouple Plate Test Setup

Problems were experienced with the combined LVDT + Thermocouple test setup. It was discovered that the A-D conversion card requires isolation of each channel. The thermocouple voltages are very low and were affected by the computer voltages.

Peak temperature readings were taken using a calibrated contact pyrometer during the balance of the test work.



3.5 Calibration of Torch Temperature and LVDT Linear Motion

The calibration of torch temperature output and LVDT'S linear motion was performed using the following detail procedures.

#### 3.5.1 Torch Calibration

Various methods of torch calibration were tried including thermal couples attached to pipe, plate and pipe rings. The vee heats were applied and the record of time/temperature was recorded. These results were then plotted using lotus 123.

These methods of torch calibration using automatic data gathering were too complex and the results were difficult to translate into a production procedure.

#### 3.5.1.1 Simplified Torch Calibration

The final torch calibration method is simple, direct and easily translatable to the shop floor.

Calibration of the torch involves setting the flow meter shown in section 3.3.1 to a preset value such as: 40 Oxygen and 35 Gas.

Both torches are ignited and the flames are held next to each other for a comparison of length and color.

The final calibration is done on a ring as shown in figure 3.5.1.1 below.

TORCH CALIBRATION RING



The above calibration is performed for each diameter and wall thickness. The calibration ring should be 4" to 6" in length.

#### 3.5.2 LVDT Linear Motion Calibration

The LVDT's will measure linear motion with accuracies of +/-.0001 inches. The LVDT type selected for this project has a measurement range of 1.0 inch. This LVDT style also measures + and - movement.

If the LVDT is used in the non calibrated node then the results a values in volts. The standard output for the system is +5VDC to -5VDC .

Calibration of any LVDT involves translating this output +/- voltage into real engineering numbers. The CODAS software has a program which works with the CODAS waveform scroller board and allows the user to gather calibration data for each LVDT (or any other recording device). See the below figure for the physical setup.



FILE: LVDT\_CAL.CGM

#### Calibration is accomplished as follows.

\* The LVDT (one LVDT per channel) is adjusted to zero on the computer screen.

\* A small segment of voltage is recorded at Zero.

\* The LVDT is repositioned to some given value. In our case a precision shim of .0925 inches thickness was used. This results in positive voltage. A small segment of voltage is recorded at this positive value.

\* The LVDT is again repositioned to a negative value using the precision shims. A final small segment of voltage is recorded at this negative value.

We now have three (3) voltage readings for one LVDT channel.

\* The final step is to run the program called CODAS POSTACQ which allows the voltage readings to be converted to engineering units. This small computer file is identified by a unique name such as HEAT\_XX.CAL.

when data is gathered in a test the HEAT\_XX.CAL file is Present and sets all test-data with the engineering units you defined. The HEAT\_XX.CAL file eliminated time consuming arithmetic errors.

3.5.3 Data Gathering System Calibration

The Key Stroke instruction detailed in APPENDIX B was used throughou calibration.

APPENDIX B can be used as a detail instruction in lieu of reading the CODAS POSTACQ instruction manual.

#### 4 LABORATORY TESTS & RESULTS

The experimental work was conducted in the Code 138 Welding Laboratory facility.

#### 4.1 VEE Heat Testing

#### 4.1.1 Temperature Measurement Techniques

A primary problem for flame bending is the accurate measurement of heat input and the maximum temperature reached during flame bending.

#### 4.1.1.1 A Simple Torch Calibration Solution

The torch calibration procedure has been previously described in section 3.5.1. A satisfactory torch calibration was considered to be when both torches heated the same spot within +/- 25° F at a given time in seconds.

The most efficient time at temperature for all 225 tests was 850° F in 15 seconds.

The standard time at temperature was derived by experimentation. The low temperature tested was  $850^{\circ}$  F in 10 seconds and the high temperature input tested was  $1000^{\circ}$  F in 15 seconds. The standard was selected as  $850^{\circ}$  F in 15 seconds.

Each pipe diameter, wall thickness and material type was tested using the 850° F in 25 second standard calibration.

#### 4.1.1.2 Maximum Pipe Temperatures

Considerable concern was expressed about the effects of maximum - temperature on the chemical and physical properties of the pipe material.

Maximum pipe temperatures were measured using a contact pyrometer on the <u>top hot spot after</u> <u>completion</u> of <u>each heat.</u> A record of the maximum temperature was recorded on the heat work record.

#### 4.1.2 The Test Matrix

The material which received the most testing was carbon steel. A typical series of tests were devised where only one test variable was changed for each series. Note that this series used a 2", 4" & 6" VEE WIDTH. No other variables were changed.

All 225 tests used the same test matrix format.

A typical test series would be as shown in the table 4.1.3

#### 4.1.3 TABLE SHOWING ONE TEST SERIES

8 Inch Diameter Carbon Steel Pipe, Schedule 40

15 Heats in ONE (1) test MATRIX

Test Parameters	Heats 36 through 40	Heats 31 through 35	Heats 26 through 30
Hinge Length of VEE			
1 Inch	36, 37, 38, 39 40	31, 32, 33, 34, 35	26, 27, 28, 29 30
3 Inch			
6 Inch			
Vee Width			
2 Inch	36, 37, 38, 39 40		
4 Inch		31, 32, 33, 34, 35	
6 Inch			26, 27, 28, 29 30

5 Heats in one test

Test heats 36, 37, 38 39 & 40 are identical heat input on the same area of pipe. The reason for running identical heats on the same location is to determine the progression of dimensional changes as they occur over the (51 five heat sequence. The second major reason is to isolate metallurgical damage, If any, to a localized-worst calocation.

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### Table 4.1.3 (Continued) 8 Inch Diameter Carbon Steel Pipe, Schedule 40

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<b>r</b>			
Test Parameters	Heats 36 through 40	Heats 31 through 35	Heats 26 through 30
Torch Tempera- ture			
850 deg / 10 sec			
850 deg / 15 sec	36, 37, 38, 39 40	31, 32, 33, 34, 35	26, 27, 28, 29, 30
1000 deg / 10 sec			
1000 deg / 15 sec		,	
Vee Spot Size			
3/4"	36, 37, 38, 39 40	31, 32, 33, 34, 35	26, 27, 28, 29, 30
1"		· · ·	
1 - 1/2"			
Vee Spot Overlap		· · · · · · · · · · · · · · · · · · ·	
1/2 Overlap		•	
Touching Spots	36, 37, 38, 39 40	31, 32, 33, 34, 35	26, 27, 28, 29, 30
1/2" Space Spots			
1" Space Spots			
Line Heats	NONE	NONE	NONE
850 deg / 10 sec			
850 deg / 15 sec			
1000 deg / 10 sec	•		
1000 deg / 15 sec			

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# 4.1.4 Relationship Between Vee Heat Pattern, Deflection and Shortening

Before proceeding with the results, a review of terms and their locations on a typical flame bending pipe may be helpful.



AFTER HEATING WITH VEE TORCH PATTERN

<u>Pipe deflection "D"</u> is defined as the lateral movement of the pipe end. Deflection <u>"D"</u> is defined as the sin of the pipe bend angle times the pipe length.

D = sin(BendPipeZ) X (PipeLength)

<u>Pipe Length</u> is defined as the distance between the Vee Hinge and the end of the pipe.

<u>Shortening</u> is defined as shrinkage of the pipe at the center line.

Shortening = 
$$\frac{EndTop + Bottornpipe}{2}$$

<u>Pipe Bend Angle</u> is self defined. This is an important measurement since it can be used to predict both deflection and flange parallel-ism.

 $\underline{\text{parallelism}}$  is defined as the sin of pipe bend angle times the diameter

D = sin(Bend PipeL) X (Pipe Diameter)

The <u>hinge length</u> we width and <u>moving spot</u> are the primary variables which control movement on pipe bends.

#### 4.1.5 Typical Time / Motion Plots

The Puget Sound Naval Shipyard is the only shipyard which currently flame bends pipe on a regular basis. Code 138 is present at each flame bending production job to assist and monitor. Most facilities rely on a highly skilled small core group of mechanics to perform this work. This core group of skilled mechanics are very good at predicting the amount of pipe movement. The emphasis in this study on motion is an attempt to build a significant data base sufficient to predict many pipe deflection questions and provide direction for future work.

Pipe movement (deflection and shortening) is information which the mechanic must have to predict what will happen when heat is applied to the pipe. Other factors of the job which are of importance to the mechanic are accessibility, pipe restraint, and safety.

# There are 225 individual plots of movement. This data are contained in APPENDIX A.

To assist the reader in understanding how a pipe moves as it is bein heated, a typical time versus movement (deflection) plot is explained in detail. The following figures are actual time - deflection curve obtained from a selected heat.

#### 4.1.5.1 Time VS Deflection (Heat\_31) 4" VEE, 1" HINGE

The X axis is numbered O through 14. This is time with each tick mark representing approximately one (1) minute. Our sample shows this time base convention. The convention on all the data in the appendix shows the time base as O to 280 which corresponds to O to 14 minutes. Our first example plot shows the time base extends to 14 minutes.

The Y axis is numbered in Thousands of an Inch. The plot is a record of movement in three axis. Four (4) transducers are used to measure X, Y, & Z with two (2) transducers on one axis. The result is four continuous line scans.



<u>THE "TOP" LINE.</u> The line marked "TOP" is a record of pipe movement 25" from the VEE on the pipe top. Notice that the pipe moves upward reaching .130" deflection within approximately 1.5 minutes due to heating below the neutral axis. At the 3 minute mark the pipe is back to 0. At 3.5 minutes the pipe is deflected to a negative -.040" due to heating above th neutral axis. THE WATER COOLING IS APPLIED TO SPEED COOLING AND THE PIPE IS TYPICALLY WELL BELOW 950° F AT THIS POINT. Th pipe begins upward movement and reaches 0 at approximately 7 minutes. At 14 minutes the pipe is at equilibrium (cold) and has a permanent deflection of .037".



 $\neg \neg$ 

THE "END TOP" LINE. The line marked "END TOP" goes negative in less than 1 minute to -.010" due to lengthening the pipe below the neutral axis. It then goes through 0 and then positive to .043" at 3.5 minutes due to overall pipe length expansion. THE WATER COOLING EFFECT TAKES PLACE and the pipe length shortens. The pipe reaches its original length at approximately 6.5 minutes and is .020" shorter at 14 minutes.



THE "BOTTOM PIPE" LINE. The line marked "BOTTOM PIPE" is the measure of movement at the pipe end bottom, on the same plane as the pipe hinge. The movement detail is read on this plot line as described in the previous "TOP" and "END TOP".

This measurement is always positive while the pipe is hot. During cooling, the measurement decreases and eventually is negative reflecting a shortened pipe. The pipe end bottom shortens less than the pipe end top reflecting a change in parallelism.



THE "SIDE PIPE" LINE. This line shows operator technique and directly reflects the effect of relative torch motion, torch position and torch temperature. If one operator holds the torch at a different distance from the pipe surface, the plot will show the result. The net result of our sample plot is that "SIDE PIPE" deflection was negative toward the operator that put more heat into one (1) vee segment. An ideal heat would be when the "SIDE LINE" returns to zero.



# 4.1.5.2 The Relationship of Hinge Size and Vee Width Versus Deflection

The following graphs illustrate how the pipe ends deflect upward as the vee width and hinge length change.

Notice that there are five (5) lines which represent each heat. The first group of data (5) are the 1" hinges, the second group c five (5) are the 3" hinges and the third group of five (5) are th 6" hinges. Also note that most data are grouped into very tight clusters which indicates that the pipe end has moved the same amount each time it was heated.

#### 4.1.5.2.1 FIXED (2") VEE SIZE with Increasing Hinge

The following graph shows two (2) important results; First th pipe material deflects approximately the same amount each time it is reheated and Second increasing hinge length results in increased deflection.





Graph 4.1.5.2.1 shows 5 heats on 8" carbon steel pipe. All parameters were held constant except the hinge length was increased on each set.

The first heat (solid line / square boxes) does not follow the other tight grouped data points. The (3) and (6) inch first heat hinge deflections may be low because of restraint by the small vee area versus the large hinge length. It has been suggested that buckling may cause the differences in data deflection after the first heat.

Heats 2, 3, 4 and 5 are verytight data sets and can be relied upon to repeat.

#### 4.1.5.2.2 FIXED (4") VEE SIZE with Increasing Hinge

The following graph shows the 3" hinge with a 3 vee to be the best combination when maximum deflection is required.



# Graph 4.1.5.2.2 shows 5 heats on 8" carbon steel pipe. All parameters were held constant except the hinge length was increased on each set.

You will note that the data is very tight on each set except the 4" vee with a 6" hinge shows spread on the 2nd heat.

There appears to be a relationship between the total area of the vee heat surface and the hinge length. The heats with very tight groups of data show a dimensionless ratio of from .005 to .03.

### $R = \frac{(\text{Hinge Length})}{(\text{TotalVeeArea})}$

This relationship could allow the user to predict a particular combination of vee size and hinge length that may deflect the same amount on each heat.
#### 4.1.5.2.3 VEE (6") FIXED Hinge Increasing

6" Vee with 1" Hinge Size 8" Carbon Steel Pipe Deflection 0.1 0.08 0 0.06 NO DATA TAKEN FOR 6" VEE 0.04 WITH 3" & 6" HINGES. ! 0.02 FILE: 6V\_136HB.CGM 0 3 INCH 6 INCH 1 INCH Hinge Length First Heat Second Heat Third Heat Fourth Heat Fith Heat

Graph 4.1.5 .2.3 shows 5 heats on 8" carbon steel pipe. All parameters were held constant and the results for heats on hinge size 3" and 6" are missing.

As in the other heats data sets are very close grouped.

#### 4.1.6 Discussion of Results on Vee Hinges

It has been shown that vee heats are predictable. There are some general relationships that apply to all vee heats on pipe. The primary parameters which control these relationships are:

Hinge length Vee width Spot temperature

Regardless of how these parameters are adjusted, the finish pipe <u>ALWAYS SHORTER AFTER HEATING.</u>

#### 4.2.6.1 Relationship of Hinge, Vee & Shortening

Short hinges + narrow vees = maximum shortening and minimum deflection.

Short hinges + wide vees = maximum shortening at the End Bottom and maximum shortening and maximum deflection.

Long hinges + narrow vees = minimum shortening and minimum deflections.

Long hinges + wide vees = zero.(0) to minimum shortening at the End Bottom and medium deflections.

Long hinges = higher residual stress, increased pipe movement and greater instability.

Wide vees = chance for increased buckling.

#### 4.1.7 Torch Travel Speed and Temperature Gradient

The average travel speed of the moving spot was in the range of 1.5 inches/minute to 3.5 inches/minute.

The overall average was 2.5 ipm. The amount of upset (movement or deflection) did not move significantly outside of a group of data points when the slower travel speed was compared to the higher trave speed.

The essential variable was spot size control and surrounding tempera ture of the support base metal of the pipe. The operators adjusted the torch standoff distance to control spot size and color. To do this they adjusted the travel speed in the range of 1.5 to 3.5 ipm.

These values were calculated from the time base on selected data plots for 8" diameter pipe.

#### 4.1.8 Determination of Residual Stress

No experiments were conducted for measurement of residual stress.

#### 4.2 Line Heat Testing

Line heats are made in using a series of straight line heating patterns with the torch moving from the hinge area to the pipe top-or from the pipe top down to the hinge area.

The tests in our study placed the lines in an area that mirrors the verheat pattern area. Figure 4.2 is a typical LINE HEAT UP pattern.



Line heats have advantages when bending pipe With very thin wall thic ness or pipe with diameters larger than 12 inches. Line heats will reduce the tendency for buckling. Vee heats will buck on thin wall and/or large diameter-thin wall pipe.

#### 4.2.1 Relationship Between Line/Vee Heat Pattern and Deflection

Line heats move in a series of discrete steps. The example shown in Figure 4.2.1 is a LINE HEAT UP PATTERN.

The first line heat causes the pipe to move down approximately .020". The second double line heat causes the pipe to move upward .040". The third double line heat causes the pipe to move upward .1". The fourth double line heat causes the pipe to move upward .070".

The combined movement of the seven (7) discrete heated lines is .190 of upward pipe movement."



Vertical Up Line Heat Patterns follow the same general pattern of pipe movement regardless of pipe diameter or material. Line heats produce more pipe movement when compared to vee heats.

#### 4.2.2 Effects of Multiple Heats

This series of heats show four (4) different materials with three (3 different pipe diameters.

ALL of these tests share a common vee size to hinge size ratio. All tests were made with a 30° vee and a corresponding hinge. Maintaining this ratio produces pipe movements with similar values. This means that the shop can predict pipe movement with a reasonable degree of certainty. The pipe range on which we now have data is from 2" to 8" diameter with Carbon Steel, Stainless Steel, 70-30 Copper Nickel and 90-10 Nickel Copper.

As the vee size to hinge ratio is changed "then the average group heights will change.



#### 4.2.3 Temperature Measurement Techniques

Line heats were monitored for maximum temperature using the same method as was used for the vee heats.

#### 4.3 Mechanical & Metallurgical Testing

One of the major objectives of this program was to determine how mechan ical and metallurgical properties are affected by the flame bending pro cess.

A simple method was used to verify and compare mechanical and metallurgical properties. The method involved selecting pipe specimens from th test matrix which received high heat. The maximum peak temperature was recorded on each test pipe. This recorded test temperature data was used to select the pipe for mechanical and metallurgical tests:

#### 4.3.1 Mechanical Test Specimen Removal Diagram



# 4.3.2 Tensile Strength, Yield Strength & Microstructure

The table 4.3.2.1 shows the relationship between heating and final mechanical properties. The specimens were removed from the BASE METAL, HIGH TEMPERATURE VEE LOCATION and the LOW TEMPERATURE START LOCATION. This removal sequence is shown in figure 4.3.1.

<sup>\*</sup>Metallographic evaluations of carbon steel, 90:10 and 70:30 CuNi and CRES are shown in Appendix B.

#### 4.3.2.1 Stainless Steel

#### PHYSICAL TEST SUMMARY For 300 Series (CRES) Stainless Steel

Sample Series 131	Tensile Load PSI	Elongation	Specification Requirements
Base Metal 131-1	86,500	64%	75,000 35%
Low Temperature 131-2	86,500	53%	75,000 35%
High Temperature 131-3	87,000	58%	75,000 35%

The tensile value differences between the original stainless steel base metal and the heat affected stainless are. less than 1%. These values are less than the calibration requirement for the Tensile Test Machine.

Elongation is related to grain size. The base metal has the highest elongation with a relatively small grain size. The heat affected tensile tests both show lower elongation with larger grain size.

The physical tests show that CRES (Stainless Steel) can be heated in the range of 1000° to 1500° F with a small reduction in elongation.

These results are well within the acceptable requirements for CRES (Stainless Steel) PIPE.

An intergranular corrosion test was conducted on the multiple heated area of flame bent Stainless Steel pipe. The test was conducted *in* accordance with ASTM 262 Practice E.

Mr. Al Ruedibusch and Mr. Mike Allen. reported "ASTM 262 intergranular corrosion test was conducted for 24 hours and the bars evaluated. There was no sign of fissures or cracks which would indicate the presence of intergranular attack." (7)

<sup>7</sup> The Puget Sound Naval Shipyard, Quality Assurance Office, Laboratory Division; Report 90Ps00424, dated January 9, 1990.

#### 4.3.2.2 Carbon Steel

#### PHYSICAL TEST SUMMARY For Carbon Steel Pipe

Sample Series 50	. Tensile Load PSI	Elongation	Specification Requirements
Base Metal 50-1	81,000	38%	60,000 22%
Low Temperature 50-2	81,000	32%	60,000 22%
High Temperature 50-3	79,500	37%	60,000 22%

The tensile value differences between the original carbons steel base metal and the heat affected carbon steel are less than 1%. These values are less than the calibration requirement for the Tensile Test Machine. (In other words the <u>differences</u> are <u>insig</u>-<u>nificant</u>)

# The carbon steel base metal has slightly higher elongation with a uniform grain size. The heat affected tensile tests exhibit very small grain size at the surface.

The resultant is a fine grain annealed heat affected region. This region is characterized by lowering of hardness with the tensile strength maintained by the fine grain structure. This fine grain indicates higher impact properties or toughness.

These results are well within the acceptable requirements for Carbon Steel Pipe.

#### 4.3.2.3 Copper Nickel (90 / 10)

#### PHYSICAL TEST SUMMARY For 90 / 10 Copper Nickel

Sample Series	Tensile	Elongation.	Specificatio
76	Load PSI		Requiremen

Base Metal 76-1	40,500	60%	38,000	30%
Low Temperature 76-2	42,600	34%	38,000	30%
High Temperature 76-3	46,000	47%	38,000	30%

The tensile values of the low and high temperature heat affected 90/10 areas are 5% to 13% greater than the as received base metal. The elongation values on both heat affected specimens are substantially less than the base metal, but still meet the specification minimums.

The 90/10 copper nickel base metal shows a large number of close packed flow lines. The heat affected pipe shows significantly fewer flow lines indicating that the segregated constituents are going into solution as a result of multiple heats.

Lack of significant grain growth is an indication that the flame bending temperatures did not exceed that of the original pipe processing annealing temperature.

These results meet the minimum acceptable requirements for 90/10 Copper Nickel Pipe.

#### 4.3.2.4 Copper Nickel (70 / 30)

#### PHYSICAL TEST SUMMARY For 70 / 30 Copper Nickel NEW PIPE

Sample Series 121	Tensile Load PSI	Specificat Elongation Requireme:		ition ents
Base Metal 121-1	57,000	31%	50,000	30%
Low Temperature 121-2	56,500	26%	50,000	30%
High Temperature 121-3	57,500	51%	50,000	30%

#### 70/30 Copper Nickel SUBMARINE MAIN SEAWATER PIPE In Service for 20 Years

Sample Series 176	Tensile Load PSI	Elongation	Specificatior Requirements	
Base Metal 176-1	57,500	28% (note 8)	50,000	30%
Low Temperature 176-2	51,900	27%	50,000	30%
High Temperature 176-3	60,500	40%	50,000	30%

The two charts above are shown together for comparison purposes. Note that the all tensile values for both the NEW PIPE and the SEA WATER SERVICE PIPE are within 10%. The major change is on the NEW PIPE which shows an increased elongation in the high temperature zone.

Three (3) of the Five (5) elongation test values are below the specification minimum requirement. However these values were measured in a region of mixed heat treatment. Values taken in these regions can be in error by +/-5%. The lowest elongation value was under the minimum by 4%.

<sup>8</sup> Note: THE NEU MANUFACTURE BASE HATERIAL ELONGATION IS BELOW THE SPECIFICATION REWIREHENT AS RECEIVED FROM THE STOCK SYSTEM.

The microstructure of 70/30 Copper Nickel test samples 121 (NEW) and 176 (Seawater) were evaluated by Metallurgists Mike Allen Al Ruedebusch. Their report, Reference 8, states that "AS seen from enclosures 4 and 5, there appears to be no significant difference between the samples taken from the base metal or the HAZ. There also appears to be no difference between the NEW 70/30 and the **SEA WATER 70/30.** (See Appendix B)

**The** net result that flame bending has minimum negative metallurgical impact on the 70/30 COPPER NICKEL pipe microstructure.

#### 4.3.3 Effect of Cooling Methods On Microstructure

The method of cooling on all 225 heats employed a circular water cooling sequence when the pipe was below 1000° F in the heated area. Pipe alloys in this study are not sensitive to rapid cooling below 1600° F. Maximum measured top temperatures were 1500° F.

No adverse metallurgical conditions were identified as a direct result of the rapid water quench.

#### 4.3.4 Material Thickness Changes

#### 4.3.5 Surface Conditions after Flame Bending

The surface conditions of all 225 heats on 5 different material types shows <u>no</u> significant surface damage.

The microphotos support the visual examinations.

#### 4.3.6 Internal Sea Water Contamination Metallurgical Results

Metallurgical examination was performed on both new 70/30 Conner-Nickel pipe and sea water service pipe. The internal surface contions after multiple flame heats were compared. No significant differences were noted by the metallurgy department.

<sup>8</sup> METALLURGICAL EVALUATION OF MICROSTRUCTURE AFTER FLAME BENDING PROCEDURE, Puget Sound Naval Shipyard, Code 134.6; Report No. 89PS05135, Dated 19 May, 1989.

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## 4.3.7 Hardness Surveys

a.

Surface hardness measurements were taken from unaffected base metal and from the heat affected zone of the pipes. Table 4.3.7 details the results.

Pipe Material Tested	Stainless Steel (CRES)	Carbon Steel (CFe)	Copper Nickel 90/10	Copper Nickel 70/30	Copper Nickel (Submarine)
	Pipe No.	Pipe No.	Pipe No.	Pipe No.	Pipe No.
	131	50	76	121	176
Base Metal	80 Rb	82 Rb	28 Rb	39 Rb	47 Rb
Heat	55 Rb .	52 Rb	18 Rb	17 Rb	26 Rb
Affected	to	to	to	to	to
Zone	71 Rb	75 Rb	22 Rb	58 Rb	50 Rb

### Hardness Survey Results Readings are all **Rockwell** B scale

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#### 5 Future Work

#### 5.1 Write Shop Working Instruction

The shop working instruction should be a document which is no longer than about 25 pages.

#### 5.2 Develop Operator Qualification Procedure

Operator qualification procedures can be written from the engineering report in cooperation with shop 56.

#### 5.3 Cooling Methods

Only one method was used on the 225 tests of this study. The method that was used in this work closely matches the method used by Puget Sound Naval Shipyard.

#### 5.4 Effects of External Restraint

The experimental work did not cover the effects of external restraint on pipe bending.

#### 5.5 Low Temperature Flame Bending

Richard E. Holt determined that the ideal flame bending temperature for most structural alloys is in the range of 700 degrees F.

Work needs to be performed to verify that low temperatures will work for pipe materials.

#### 5.6 Thickening in the Flame Bent Area

More accurate values of thickening should be made in the hot region.

#### 5.7 Charpy Impact Values

Limitedwork has shown impacts are improved on some materials but additional work needs to be performed on a broad range of naval materia

<sup>9</sup> Holt, R. E., "PRIMARY CONCEPTS FOR FLAME BENDING", Weiding Journal, June 1971.

#### 6 CONCLUSIONS

#### 6.1 Precise & Simple Torch Calibration

A method of oxyacetylene torch heat calibration has been developed using precision flow meters and test ring.

#### 6.2 Vee Heating Pattern VS Line Heating Patterns

Vee heats are more predictable for smaller diameter pipe in the range of 2" diameter to 8" diameter. Vee heats should be used as a first choice by the shop.

The majority of-naval and commercial shipyard pipe flame bending can be performed using 3" VEE HEATS. This VEE size will produce significant deflection with minimum buckling.

#### 6.3 Line Heats on Large and/or Thin Wall Pipe

Line heats were found to be more controllable on pipe diameters larger than 10 inches. Pipe with very thin wall in relation to its diameter should be flame bent using the line heat method.

Line heats with a given surface heated area produce greater deflection than the equivalent vee heat with the same heated surface area.

#### 6.4 Dimensionless Ratio for Hinge / Vee Area

There is a relationship between the hinge length and total surface area of the vee heat. When the above two values yield a number from .005 to 03 then the mechanic can assume that each heat sequence will deflect the pipe the same amount on each heat.

#### R - <u>CHingeLength</u>) CTotaLVeeArsa)

The mechanic can reheat pipe material in the same area repeatedly and obtain predictable movement.

#### 6.5 Ideal Travel Speed

The ideal travel speed for most small diameter flame bending application is in the range from 1.5 to 3.5 ipm.

#### 6.6 Maximum Number of Heats on the Same Area

It has been determined by experimentation that no more than three (3) heats should be applied to the same area. This maximum is not a metallurgical limit but is determined by the pipe materials tendency to buckle after the third heat.

#### 6.7 Pipe Movement On Carbon Steel

Carbon steel moves approximately the same amount (with a given heated area) each time it is heated in the same location.

#### 6.8 Stainless Steel

The physical tests show that CRES (Stainless Steel) can be heated in the range of 1000° to 1500° F with a small reduction in elongation and limited metallurgical damage. ASTM 262 intergranular corrosion tests were conducted with no sign of fissures or cracks.

#### 6.9 Carbon Steel Metallurgical Property Improvement

Carbon steel showed generally improved overall physical properties.

#### 6.10 Copper Nickel Metallurgical Properties after Flame Bending

Flame bending produces minimum metallurgical damage on the 70/30 COPPER NICKEL pipe microstructure.

#### 6.11 Effect of Flame Bending on Sea Water Pipe

There is no measurable metallurgical or mechanical property difference between the NEW 70/30 and the SEA WATER CONTAMINATED 70/30 as a result of flame bending.

There is a slight reduction in Tensile, Yield strength and elongation on both new and sea water flame bent pipe: The reduction is less than 1%.

#### 6.12 Dye Penetrant Testing

Each material type was PT tested after five heats and no evidence of cracking was found on any of the material tested.

#### 6.13 Scaling and Flaking (internal)

90/10 copper nickel shows measurable internal scaling and flaking after flame bending. It is recommended that flame bending on 90/10 not be performed unless purged during bending or flushed after bending. This should apply to all FEMA and hydraulic piping.

All other materials tested showed discoloration but did not scale.

#### 6.14 Flow Meter Calibration

A flow meter may be substituted as the primary calibration method once the initial ring/flame length calibration is performed. Individual flow meters will significantly reduce out of balance heat input.

#### 6.15 Larger Hinge Lengths

Tests show that larger hinge lengths in the range of 4" to 5" on 5" to 8" pipe yield controllable deflections. Current methods use 2" hinge lengths which are too small for most applications.

#### 6.16 Optimum Average Vee and Hinge Ratios

Most shop work can be accomplished with a 30° vee angle and hinge which is 1/4 or (25%) of the circumference:

#### 6.17 When to Use Longer Hinges

Longer hinges should be used when pipe shortening must be held to a minimum.

#### 6.18 Sea Water Pipe Bends

70/30 sea water pipe was tested using a standard AWS B4.0-77 guided bend test. There was no indication of cracking. Dye penetrant tests on this material before bending showed no indications.

#### 6.19 Trained Mechanics

Do not use personnel who have no formal flame bending training. Operators should train on similar grades, diameters and wall thicknesses.

# 6.20 Flame Bending of Stainless Steel (CRES)

Based on the test work Stainless steel can be flame bent with no significant loss of mechanical or metallurgical properties.

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# APPENDIX A

CODAS Calibration Data Gathering Method	
To start Data Gathering to Hard Disk	A-1
To stop Data Gathering to Hard Disk	A-1
To Display Data Gathering during Flame Bending Test	A-1
To Display the eight (8) Data Screen	A-1, A-2

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#### CODAS CALIBRATION DATA GATHERING METHOD

ENTER SELECTION : (ie) 6 filename [with no extension]

SET COLOR MONITOR: <CTL> <ALT> <BACK ARROW> [monitor is now showing a display.

#### TO START DATA GATHERING TO HARD DISK

<F10> [monitor is now showing byte count in lower right
hand corner]

#### TO STOP DATA GATHERING TO HARD DISK

<F9> [monitor is now showing final byte count in lower right corner]

<q> TO RETURN TO DOS</q>	[monitor shows prompt]
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#### TO DISPLAY DATA GATHERED DURING FLAME BENDING TEST

TYPE IN: POSTACQ filename.ext [postacq is the name of the data display and manipulation program. The file name is the name given to the initial flame bend test.]

#### TO DISPLAY TEE 8 DATA SCREENS

<shift o=""></shift>	
<f1></f1>	[enable the data cursor]
< <> tion]	> [position the cursor to the desired posi-
<f4></f4>	[enable the final data gathering point]
<f6></f6>	[enables end of data gathering sequence]

SCREEN SHOWS: ENTER THE NAME OF DATA FILE:

Enter any name that you want to use to identify this particular data record. This file may be the whole test or part of the test sequence.

Note: If the data is going to be used in LOTUS then an extension of .WK1 should be used.

If the data is going to be printed directly, then use an extension of .HCU.

ENTER FILE NAME: <heat\_X.wkl>

MENU APPEARS: [select #5 or #6] INPUT TEXT: [two lines of text can be input at this point]

#### HIT <RETURN>

WINDOW DISPLAY RETURNS: [hit <Q> to quit. CALIBRATION CONSTANTS: [y/n] BACK TO DOS:

#### APPENDIX "B"

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**D**.

APPENDIX "B"	Metallurgical Evaluation of Microstructure after Flame Bending	Page
	Hardness Test Results	B-1
	Microstructure Evaluation	B-2
	Tensile Test Results after Flame Bend	B-3
	Microstructure of Stainless Steel Pipe before and after Flame Bending	B-4
	Microstructure of Carbon Steel Pipe before and after Flame Flame Bending	в-5
	Microstructure of 90:10 Copper Nickel before and after Flame Bending	в-6
	Comparison of Microstructure of Bare Metal and HA2 Micro- Structure for CuNi 70:30 Pipe with Sea water Service and New Pipes	B-7 and B-8

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# PUGET SOUND NAVAL SHI PYARD QUALI TY ASSURANCE OFFI CE Laboratory Division 19 May 1989

Code 134.6

MFA: I ms

Metallurgy Branch

Report No. 89PS05135

METALLURGICAL EVALUATION OF MICROSTRUCTURE AFTER FLAME BENDING Subj : PROCEDURE

(a) American Society for Testing and Materials, <u>Annual Book of ASTM</u> <u>Standards</u>, Volume 1.05, Designation A 262-86, 1987 Ref:

- Photomi crographs of Pi pe 131
   Photomi crographs of Pi pe 50
   Photomi crographs of Pi pe 76
   Photomi crographs of Pi pe 121
   Photomi crographs of Pi pe 176 Encl:

1. Five samples were provided by Code 138 for hardness testing and evaluation of the microstructure. The samples were taken from sections of pipe that had been cycled through a flame bending procedure. The cycle consisted of heating from room temperature to approximately 1500°F and then water quenching. There were a total of five cycles.

2. <u>Hardness</u>. Surface hardness measurements were taken from the unaffected base metal and from the heat affected zone of the pipes, see table below.

Results of the Hardness Transverse					
Area Tested	131 Cres New	50 CFe New	76 90/10 New	121 70/30 New	176 70/30 *
Base Metal (Ave)	80 HRB	82 HRB	28 HRB	39 HRB	47 HRB
HAZ (Range)	71 to 55	75 to 52	22 to 18	58 to 17	50 to 26

\*Pipe sample was obtained from an inactive boat and had been exposed to sea water.

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3. <u>Microstructure</u>. Two samples were sectioned from each pipe to determine what effects the flame bending procedure had on the microstructure, if any. One sample was taken from the base metal and one from the heat affected zone.

Pipe 131:

Due to stainless steel's susceptibility to sensitization, an oxalic acid etch test was performed on this sample. ASTM Standard A 262, Practice A, was used to determine what degree of sensitization the flame bending procedure produced in the steel. As seen in enclosure (1), there is a difference between the sample from the HAZ and the base metal. The base metals figure 1, shows a step structure while the sample from the HAZ, figure 2, has a dual structure with some ditching. This indicates that there is some sensitization which increases its susceptibility to intergranular attack. While Practice A cannot be used to reject the material, it does show that more testing is required.

#### Pipe 50:

Enclosure **2**, figures **1** and **29** shows that the flame bending Procedure has a recrystallization effect on the carbon steel. The sample from the HAZ has smaller grains near the surface, whereas the sample from the base metal has uniform grain size throughout.

Pipe 76:

The flame bending cycle, for the 90/10, appears to alter the flow pattern of the material. As seen in enclosure (3), figure 1, there are a large number of flow lines present in the base metal sample. Figure 2 is from the HAZ and the number of flow lines have been significantly reduced. There is no indication of recrystal-lization-or grain growth.

Pipes 176 and 121:

As seen from enclosures 4 and 53 there appears to be no significant difference between the samples taken from the base metal or the HAZ. There also appears to be no difference between the new 70/30 and the sea water 70/30.

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4. <u>Tensile Tests</u>.

TENSILE TEST RESULTS						
SAMPLE	UTE KSI	<pre>% ELONG.</pre>	FAILED GAUGE Y	INSIDE MARKS N	REQ'D KSI	REQ'D ELONG.
131-1 131-2 131-3	86.5 86.5 87.0	64 53 58	X X X		75.0	35%
50-1 50-2 50-3	81.0 81.0 79.5	38 32 37	X X X		60.0	22%
76-1 76-2 76-3	40.5 42.6 46.0	60 34 47	X X X		38.0	30%
121-1 121-2 121-3	57.0 56.5 57.5	31 26* 51	x	x x	50.0	30%
176-1 176-2	57.5 51.9	28* 27*		x x	50.0	30%

\*These values are below the accepted minimum.

I Fallen <u>Metallurgist</u>

Head, Metallurgy Branch

Distribution: Code 138(2 copies)

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Figure 1, Base Metal Etch: Oxalic 6A/cm<sup>2</sup> 500x



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Figure 1, Base Metal Etch: 2% Nital 100x



Figure 2, HAZ Ftch: 2% Nital 100x



Figure 1, Base Metal Etch: 10% NaCN, HOCOCOOH 100x





Figure 2, 121 HAZ

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Figure 1, Base Metal Etch: 10% NaCN, HOCOCOOH 100x



Figure 2, HAZ

#### APPENDIX C

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# TIME\* VS PIPE DEFLECTION CHARTS

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\*NOTE: On Time Scale 20 minor divisions equals 1 minute i.e. 0 - 280 = 14 minutes

Cu-Ni 90:10, 8" PIPE, 5/16" WALL

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CHART #	HEAT #	TECHNIQUE	PACE
16	1 <u>st</u>	4" Vee 6" Hinge	1
17	2 <u>d</u>		1
18	3 <u>d</u>		2
19	4 <u>th</u>		2
20	5 <u>th</u>		3
20	1 <u>st</u>	2" Vee 3" Hinge	3
22	2 <u>nd</u>		4
23	3 <u>rd</u>		4
25	5 <u>th</u>		5
26	1 <u>st</u>	6" Vee l" Hinge	6
27	2 <u>nd</u>		6
28	3 <u>rd</u>		7
29	4 <u>th</u>		7
30	5 <u>th</u>		8
31	1 <u>st</u>	4" Vee l" Hinge	8
32 .	2 <u>nd</u>		9
33	3 <u>rd</u>		9
34	4 <u>th</u>		10
36	1 <u>st</u>	2" Vee l" Hinge	10
37	2 <u>nd</u>		11
39	4 <u>th</u>		12
40	5 <u>th</u>		12
41	1 <u>st</u>	4" Vee 3" Hinge	13
43	3 <u>rd</u>		123
44	4 <u>th</u>		14
46	1 <u>st</u>	6" Vee 3" Hinge	14
47	2 <u>nd</u>		15
48	3 <u>rd</u>		15
49	4 <u>th</u>		16
50	5 <u>th</u>		16

CHART #	HEAT #	TECHNIQUE	PAGE
51	lst	2" Vee 6" Hinge	17
52	2nd		17
53	3 <mark>rd</mark>		18
54	$4\overline{th}$		18
55	5 <u>th</u>		19
FC	1 -+	All Moo fl Hingo	10
50	1 <u>80</u>	4 vee 6 Hinge	19
52	320		20
59	$4 \pm h$		20
60	5th		21 & 2

# Cu Ni 90:10, 8" PIPE, 3/16" WALL

CHART # 66 67 68 69 70	HEAT # 1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u> 5 <u>th</u>	TECHNIQUE 4" Vee 6" Hinge	PAGE 22 23 23 24 24
71	1 <u>st</u>	4" Vee 6" Hinge	25
72	2 <u>nd</u>		25
73	3 <u>rd</u>		26
74	4 <u>th</u>		26
75	5 <u>th</u>		27
76	1 <u>st</u>	2" Vee 6" Hinge	27
77	2 <u>nd</u>		28
78	3 <u>rd</u>		28
79	4 <u>th</u>		29
80	5 <u>th</u>		29
81	4 <u>th</u>	Line heat, 6" Hinge	30
82	4 <u>th</u>		30
83	4 <u>th</u>		31
84	4 <u>th</u>		31
811	l <u>st</u> Li	ne Heat Uphill, 6" Hinge	32 .
812	2 <u>nd</u>		32
813	3 <u>rd</u>		33
814	4 <u>th</u>		33
821	l <u>st</u> Li	ne Heat Uphill, 6, Hinge	34
822	2 <u>nd</u>		34
823	3 <u>rd</u>		35
824	4 <u>th</u>		35

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#### Continuation 2

APPENDIX C

CHART # 831 832 833 834	HEAT # 1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u>	TECHNIQUE Line Heat Uphill, 6" Hinge	PAGE 36 36 37 37
841	1 <u>st</u>	Line Heat Uphill, 6" Hinge	38
842	2 <u>nd</u>		38
843	3 <u>rd</u>		39
844	4 <u>th</u>		39
851	l <u>st</u>	Line Heat Uphill, 6" Hinge	40
852	2 <u>nd</u>		40
853	3 <u>rd</u>		41
854	4 <u>th</u>		41
861	1 <u>st</u>	Line Heat, Downhill, 6" Hînge	42
862	2 <u>nd</u>		42
863	3 <u>rd</u>		43
864	4 <u>th</u>		43
871	1 <u>st</u>	Line Heat, Downhill, 6" Hinge	44
872	2 <u>nd</u>		44
873	3 <u>rd</u>		45
-874	4 <u>th</u>		45
881	1 <u>st</u>	Line Heat, Downhill, 6" Hinge	46
882	2 <u>nd</u>		46
883	3 <u>rd</u>		47
884	4 <u>th</u>		47
891	1 <u>st</u>	Line Heat, Downhill, 6" Hinge	48
892	2 <u>nd</u>		48
893	3 <u>rd</u>		49
894	4 <u>th</u>		49
Ht 91	1 <u>st</u>	4" Vee, 6" Hinge, Downhill	50
Ht 92	2 <u>nd</u>		50
Ht 93	3 <u>rd</u>		51
Ht 94	4 <u>th</u>		51
Ht 95	5 <u>th</u>		52
901	1 <u>st</u>	Line Heat, Downhill, 6" Hinge	52
902	2 <u>nd</u>		53
903	3 <u>rd</u>		53
904	4 <u>th</u>		54
962	2 <u>nd</u>	Line Heat, Uphill, 12" Hinge	54
963	3 <u>rd</u>		55
972	2 <u>nd</u>	Line Heat, Uphill, 12" Hinge	55
973	3 <u>rd</u>		56

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CHART # 982 983	HEAT # 2 <u>nd</u> 3 <u>rd</u>	TECHNIQUE Line Heat, Uphill, 12" Hinge	PAGE 56 57
992 993	2 <u>nd</u> 3 <u>rd</u>	Line Heat, Uphill, 12" Hinge	57 58
1002 1003	2 <u>nd</u> 3 <u>rd</u>	Line Heat, Uphill, 12" Hinge	· 58 59
70:30 Cu-N	Ni, 8" PIPE,	<u>3/16" Wall</u>	
CHART # 101 102 103 104 105	HEAT # 1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u> 5 <u>th</u>	TECHNIQUE 4" Vee, Uphill, 6" Hinge	PAGE 59 60 60 61 61
1061 1062 1063 1064	1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u>	Line Heat, Uphill, 6" Hinge	62 62 63 63
1071 1072 1073 1074	1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u>	Line Heat, Uphill, 6" Hinge	64 64 65 65
1081 1082 1083 1084	1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u>	Line Heat, Uphill, 6" Hinge	66 66 67 67
1091 1092 1093 1094	1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u>	Line Heat, Uphill, 6" Hinge	68 68 69 69
1101	1 <u>st</u>	Line Heat, Uphill, 6" Hinge	70
1103	3 <u>rd</u>		70
CARBON STI	CEL PIPE, 8'	' NPS, 5/16" Wall	
CHART # 1111 1112 1113 1114	HEAT # 1 <u>st</u> 2 <u>nd</u> 3 <u>rd</u> 4 <u>th</u>	TECHNIQUE Line Heat, Uphill, 6" Hinge	PAGE 71 71 72 72

Continuation 3

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APPENDIX C

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Continuation 4

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APPENDIX C

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CHART #	HEAT 🛱	TECHNIQUE	PAGE
1121	l <u>st</u>	Line Heat, Uphill, 6" Hinge	73
1123	3 <u>rd</u>		74
1124	4 <u>th</u>		74
1131	1 <u>st</u>	Line Heat, Uphill, 6" Hinge	75
1132	2 <u>nd</u>		75
1134	3 <u>rd</u>		76
1135	4 <u>th</u>		76
1141	l <u>st</u>	Line Heat, Uphill, 6" Hinge	77
1142	2 <u>nd</u>		77
1143	3 <u>rd</u>		78
1144	4 <u>th</u>		78
1151	1 <u>st</u>	Line Heat, Uphill, 6" Hinge	79
1152	2 <u>nd</u>		79
1153	3 <u>rd</u>		80
1154	4 <u>th</u>		80

\*

# 70:30 Cu Ni, 5" NPS PIPE, 1/4" WALL

CHART # 121 122 123 124 125	HEAT # 1st 2nd 3rd 4 <u>th</u> 5 <u>th</u>	TECHNIQUE 2.5" Vee, 3.75" Hinge, Uphill	PAGE 81 82 82 83
1261	1 <u>st</u>	Heat, UPhill 2.5" Vee, 3.75 Hinge	83
1262	2 <u>nd</u>		84
1263	3 <u>rd</u>		84
1301	1 <u>st</u>	Line Heat, Uphill, 3.75" Hinge	85
1302	2 <u>nd</u>		85
1303	3 <u>rd</u>		86

# STAINLESS STEEL PIPE, 5" NPS, 1/4" WALL

CHART #	HEAT #	TECHNIQUE	PAGE
131	lst	2.5" Vee, 3.75" Hinge, Uphill	86
132	2nd		87
133	3nd		87
134	4th		88
135	5 <u>th</u>		88

#### APPENDIX C

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## Continuation 5

CHART #	HEAT #	TECHNIQUE	PAGE
151	lst	2" Vee, 1.75" Hinge	89
152	2nd		89
153	3rd	•	90
154	$4\overline{th}$		90
155	5th		91

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8" PIPE, 5/16" WALL





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8" PIPE, 5/16" WALL 2" VEE, 1" HINGE 0.150 -0.140 -0.130 -0.120 -0.110 0.100 FILE: HEAT\_36D.CCM 0.090 THOUSANDS OF AN INCH 0.080 -0.070 0.060 0.050 0.040 0.030 0 0.020 0.010 Δ 0.000 -0.010 -0.020 -0.030 -0.040 -0.050 -0.050 200 240 280 80 120 160 ò 40 TIME ( FIRST HEAT, FILE: Heat\_36 ) ---- BOTTOM PIPE - TOP ---- SIDE PIPE

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8" PIPE, 5/16" WALL





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2" VEE, 6" HINGE 0.150 0.140 0.130 FILE: HEAT\_52D.CGM 0.120 -0.110 0.100 0.090 THOUSANDS OF AN INCH 0.080 0 0.070 0.060 0.050 0.040 0.030 0.020 0.010 0.000 -0.010 Δ -0.020 -0.030 -0.040 -0.050 -0.060 -Ô 40 80 120 160 200 240 280 TIME ( 2nd HEAT, FILE: Heat\_52 ) O\_ TOP BOTTOM PIPE ---- END TOP

8" PIPE, 5/16" WALL



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8" PIPE, 5/16" WALL







8" PIPE, 5/16" WALL



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8" PIPE, 5/16" WALL



8" PIPE, 3/16" WALL 4" VEE, 6" HINGE 0.200 Δ 0.180 0.160 0.140 FILE: HEAT\_66D.CGM 0.120 . THOUSANDS OF AN INCH 0.100 0.080 0.060 0.040 0.020 0.000 ū -0.020 -0.040 -0.060 -0.080 40 200 80 120 160 240 280 0 TIME ( 1st HEAT, FILE: Heat\_66 ) <u>•</u> TOP 

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8" PIPE, 3/16" WALL, CU-NI







8" PIPE, 3/16" WALL, CU-NI



8" PIPE, 3/16" WALL, CU-NI

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8" PIPE, 3/16" WALL, CU-NI

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8" PIPE, 3/16" WALL, CU-NI

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8" PIPE, 3/16" WALL, CU-NI



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8" PIPE, 3/16" WALL, CU-NI

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8" PIPE, 3/16" WALL, CU-NI



8" PIPE, 3/16" WALL, CU-NI



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8" PIPE, 3/16" WALL, CU-NI FOURTH LINE HEAT, UPHILL, 6" HINGE 0.150 0.140 -0.130 -FILE: HEAT834D.CGM 0.120 0.110 0.100 0.090 0 THOUSANDS OF AN INCH 0.080 0.070 0.060 0.050 0.040 0.030 0.020 0.010 0.000 -0.010 -0.020 -0.030 -0.040 -0.050 -0.060 Ó 40 80 120 160 200 240 280 TIME ( 4th HEAT, FILE: Heat\_834) BOTTOM PIPE O TOP 















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8" PIPE, 3/16" WALL, CU-NI 4th LINE HEAT, DOWN HILL, 6" HINGE 0.150 0.140 0.130 FILE: HEAT864D.CGM 0.120 0.110 0.100 0.090 THOUSANDS OF AN INCH 0.080 0.070 0.060 0.050 0.040 0 0.030 0.020 n 0.010 0.000 -0.010 -0.020 -0.030 -0.040 -0.050 -0.060 Ó 40 80 120 160 200 . 240 280 TIME ( 4th LINE HEAT, FILE: Heat\_864 ) O TOP 



8" PIPE, 3/16" WALL, CU-NI SECOND LINE HEAT, DOWN HILL, 6" HINGE 0.150 0.140 0.130 FILE: HEAT872D.CGM 0.120 0.110 0.100 0.090 THOUSANDS OF AN INCH 0.080 0.070 0.060 0.050 0.040 0.030 0.020 0.010 0.000 -0.010 -0.020 -0.030 -0.040 -0.050 -0.060 160 200 240 280 80 ٥ 40 120 TIME ( 2nd HEAT, FILE: Heat\_872) \_\_\_\_\_ END TOP D TOP воттом ріре 

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FOURTH LINE HEAT, DOWN HILL, 6" HINGE 0.150 0.140 -0.130 -FILE: HEAT874D.CGM 0.120 0.110 0.100 0.090 THOUSANDS OF AN INCH 0.080 0.070 0.060 0.050 0.040 0 0.030 0.020 0.010 -0.000 · -0.010 ۵ -0.020 --0.030 --0.040 -0.050 -0.060 · ò 40 80 120 160 200 240 280 TIME ( 4th HEAT, FLE: Heat\_874) <u>о</u> <sub>тор</sub> A. SIDE PIPE

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8" PIPE, 3/16" WALL, CU-NI





8" PIPE, 3/16" WALL, CU-NI



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8" PIPE, 3/16" WALL, CU-NI



8" PIPE, 3/16" WALL, CU-NI





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8" PIPE, 3/16" WALL, CU-NI





8" PIPE, 3/16" WALL, CU-NI



8" PIPE, 3/16" WALL, CU-NI

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**C** 56











8" PIPE, 3/16" WALL, 70/30 CU-NI





8" PIPE, 3/16" WALL, 70/30 CU-NI

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8" PIPE, 3/16" WALL, 70/30 CU-NI



8" PIPE, 3/16" WALL, 70/30 CU-NI







8" PIPE, 3/16" WALL, 70/30 CU-NI



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8" PIPE, 3/16" WALL, 70/30 CU-NI



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8" PIPE, 3/16" WALL, 70/30 CU-NI FOURTH LINE HEAT, UPHILL, 6" HINGE 0.150 0.140 0.130 -FILE: HT\_1074D.CCM 0.120 -0.110 0.100 0.090 THOUSANDS OF AN INCH 0.080 0.070 0.060 · D 0.050 0.040 0.030 -0.020 0.010 0.000 -0.010 -0.020 -0.030 --0.040 -0.050 -0.060 -0 40 80 120 160 200 240 280 . D TOP SIDE PIPE TIME ( 4th HEAT, FLE: Heat\_1074B) BOTTOM PIPE



8" PIPE, 3/16" WALL, 70/30 CU-NI



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8" PIPE, 3/16" WALL, 70/30 CU-NI

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8" PIPE, 3/16" WALL, 70/30 CU-NI





8" PIPE, 3/16" WALL, 70/30 CU-NI

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8" PIPE, 3/16" WALL, 70/30 CU-NI

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8" PIPE, 3/16" WALL, 70/30 CU-NI

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8" PIPE, 5/16" WALL, CARBON STEEL





8" PIPE, 5/16" WALL, CARBON STEEL

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8" PIPE, 5/16" WALL, CARBON STEEL




8" PIPE, 5/16" WALL, CARBON STEEL

[PICTURE]

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8" PIPE, 5/16" WALL, CARBON STEEL

8" PIPE, 5/16" WALL, CARBON STEEL



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8" PIPE, 5/16" WALL, CARBON STEEL

8" PIPE, 5/16" WALL, CARBON STEEL

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8" PIPE, 5/16" WALL, CARBON STEEL .

8" PIPE, 5/16" WALL, CARBON STEEL





8" PIPE, 5/16" WALL, CARBON STEEL

8" PIPE, 5/16" WALL, CARBON STEEL



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8" PIPE, 5/16" WALL, CARBON STEEL





8" PIPE, 5/16" WALL, CARBON STEEL





8" PIPE, 5/16" WALL, CARBON STEEL







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5" PIPE, 1/4" WALL, 70/30 CU-NI



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5" PIPE, 1/4" WALL, STAINLESS







5" PIPE, 1/4" WALL, STAINLESS





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2" PIPE, .352" WALL, MONEL, NI-CU



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