Dr. Steven H. Walker Program Manager DARPA/TTO

#### **Common Aero Vehicle**

High velocity payload delivery system

> Ken Qassim AFRL/VS

Joint DARPA/Air Force Force Application and Launch from CONUS Technology Demonstration

Hypersonic Cruise Vehicle Prompt global reach

#### Small Launch Vehicle

Operationally responsive and affordable spacelift

Dr. Russ Partch Project Manager AFRL/VSE

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# PROGRAM GOAL



Develop and Validate, In-flight, Technologies that will Enable Both Nearterm and Far-term Capabilities to Execute Prompt Global Strike Missions while at the Same Time, Demonstrating Affordable and Responsive Space Lift

# HYPERSONIC FORCE APPLICATION AND LAUNCH TECHNOLOGY DEMONSTRATION

# FALCON Program Advances Technology Necessary for Prompt Global Strike Capability Prompt Global Strike (PGS) Requirements Strike globally and rapidly with joint forces against high-payoff targets In a timeframe reduced from weeks/days to hours/minutes Even when no permanent military presence or only limited infrastructure is in a region Regardless of anti-access threats

>In a single or multi-theater environment

Weapons of Mass Radiological Destruction Chemica iological Advanced Nuclea Conventional Theater-Weapons Range Missiles and-Attack Cruise Missiles 'Double-Diait' Hard and Deep C2 Bunkers SAMs Targets Decentralized, WMD Storage Internetted IADS 
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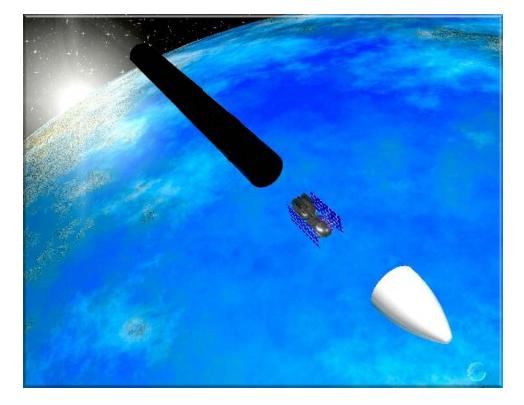
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# SMALL SATELLITE LAUNCH





# <u>Operationally</u> <u>Responsive Spacelift</u> <u>Capability</u>

» Small ISR payloads to Sun
 Synchronous Orbits

- » Low Recurring Launch Cost
- » New Launch OperationsParadigm

# FALCON Hypersonic Technology Program will Achieve Near and Far Term PGS Objectives



# Near Term Operational System

Common Aero Vehicle (CAV) and Small Launch Vehicle (SLV) System Capability: >High Endurance CAV

- Multiple payloads
- Multiple munitions
- > Operationally responsive booster
- ≻Global range
- Extended cross range

# **FAR Term Operational System**

Hypersonic Cruise Vehicle (HCV) System Capability:

- High Lift/Drag Configuration
- Multiple use payload bays
- Global down and cross range
- Aircraft operations
  - Reusable
  - Recallable
  - Launch on demand

Distribution: Gov & Gov Contractors, ITAR Restricted



# Common Aero Vehicle (CAV)



**Objective: CAV Technology Demonstration Flight Test** 

Description of CAV:
Lifting aeroshell surviving >Mach 22 reentry velocity
Maneuverable: >3000 nm range and >1000 nm crossrange
Controllable with 3m accuracy objective

*Hypersonic Technology Vehicle Flight Experiment:* > Prototype vehicle with available (SOTA) technology

- Vehicle Test flight with instrumentation (no weapons)
  - TPS effectiveness & endurance (3500 F outside, 160 F inside)
  - Integrated aerodynamic performance
  - Guidance, navigation, and control
  - Communications and plasma effects

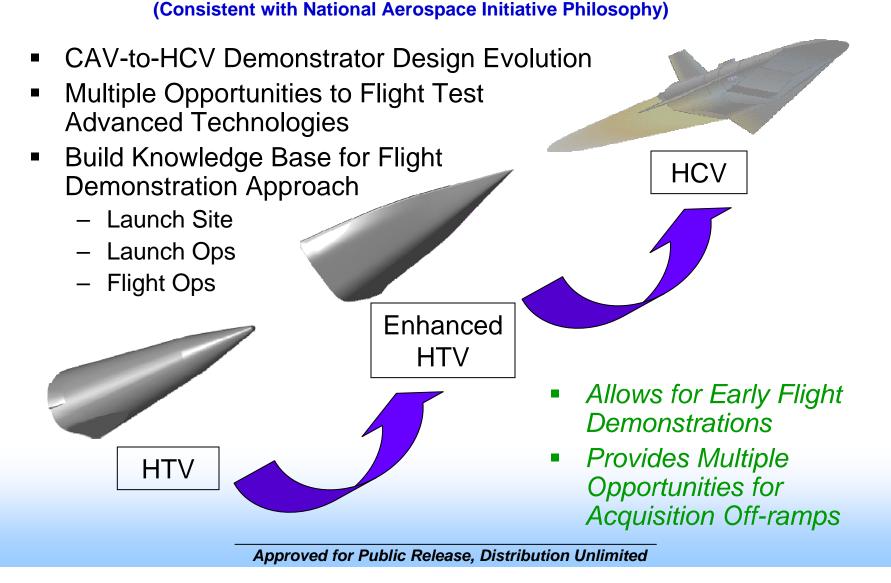
# DARPA

## HYPERSONIC TECHNOLOGY EVOLUTION



7

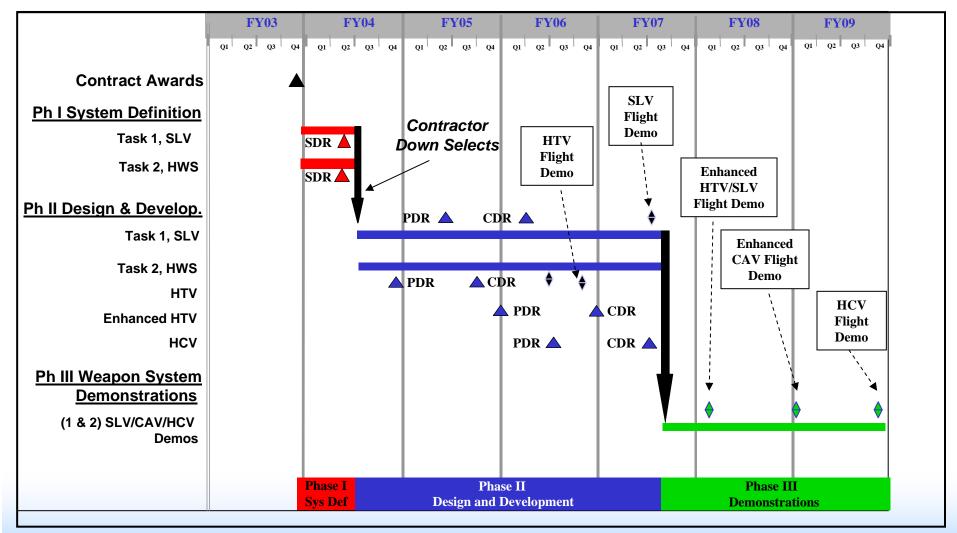
Building Block Tech Development and Flight Demo Approach





# **PROGRAM SCHEDULE**





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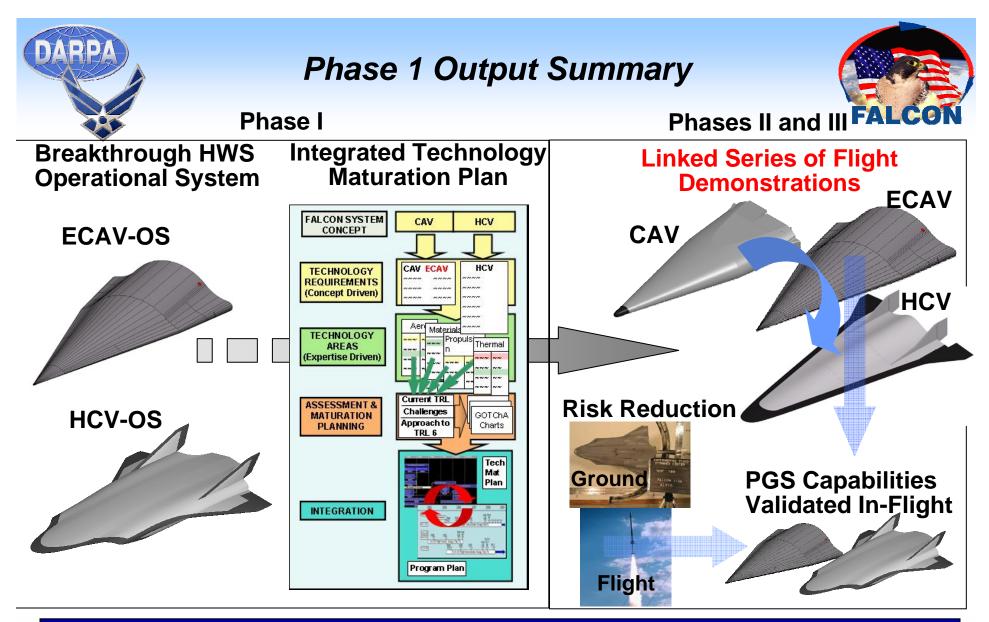
LOCKHEED MARTIN /



Lockheed Martin selected for hypersonic technology development and flight demonstrations in phase II of the Falcon program.







Our in-flight demonstration program enables adoption of hypersonic prompt global strike solutions by the warfighter

#### FALCON CAV Operational System Meets Near Term Prompt Global Strike Objectives



Small Launch Vehicle System

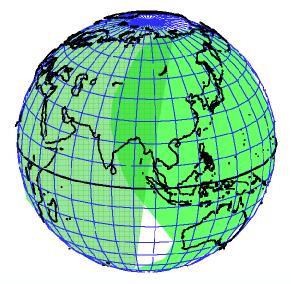
Advanced CAV Aero-shell and Insulation



**Multiple Payload Carriage** 

High Lift/Drag Advanced GN&C and Communications

Enables CAV Global Reach from CONUS



Terminate and Re-target Capability

Operation CAV/SLV System provides the warfighter with transformational prompt, precision worldwide strike capability from CONUS

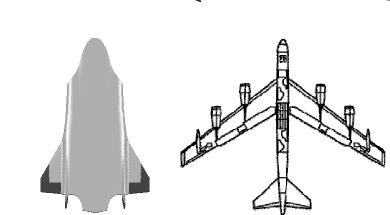
#### FALCON HCV Operational System Meets Far Term Prompt Global Strike Objectives



**Technology Advances Required for:** 

- Aerodynamic Vehicle Design
- Inward Turning Propulsion System Integration
- Passive Small Radius Leading Edges
- Metallic Encapsulated Thermal Protection System
- Hot and Warm Structure Technologies
- Internal Cryogenic Insulation
- Conformal Tanks
- Mixed Phase Hydrogen Pumps

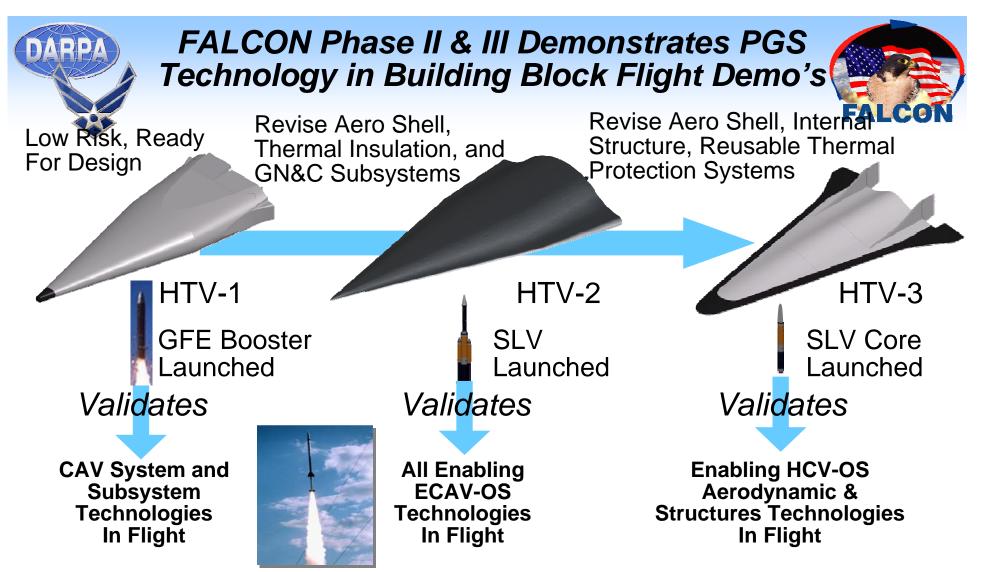
FALCON HCV can strike the depth of any adversaries territory at a size and cost acceptable to the warfighter



**B-52 Size and Weight Class** 

#### FALCON Program Will Demonstrate **Operational System Enabling Technologies** 2005 2007 2008 2009 2004 2006 **Phase** Phase II Phase III PDR Flight HTV-1 Design Fab & Integration $\nabla P \mathbf{D} \mathbf{R}$ CDR Flight 1 HTV-2 Design Flight 2 Fab & Integration Fab & Flight Test 7 PDR CDR Flight 1 HTV-3 Demonstrator Design Flight 2 Flight 3 Fab & Integration Flight Test Technology Risk Reduction Inward Turning Scramjet Flight Test **Operational Systems Refinement**

Three Distinct Hypersonic Technology Vehicles (HTV) Focus Technology Maturation in a Building Block Approach



Inward Turning Engine Flight Demonstration Validates Enabling HCV-OS Propulsion Technologies in Flight

> FALCON's evolutionary, spiral development flight demonstrator approach reduces technology validation cost and risk



# **HTV-1** Demonstration System Summary

HTV-1 uses state-of-the-art materials and

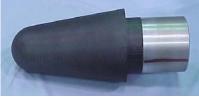
components to reduce overall program risk

and demonstrate today's Common Aero

Vehicle hypersonic technology capability

HTV-1





Nosetip



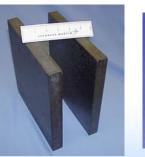
Aft Cover



**IR&D** Aeroshell



**Antenna Window** 



**Carbon-carbon** samples



**ESIGI** 



Encoder

LOCKNEED MARTIN

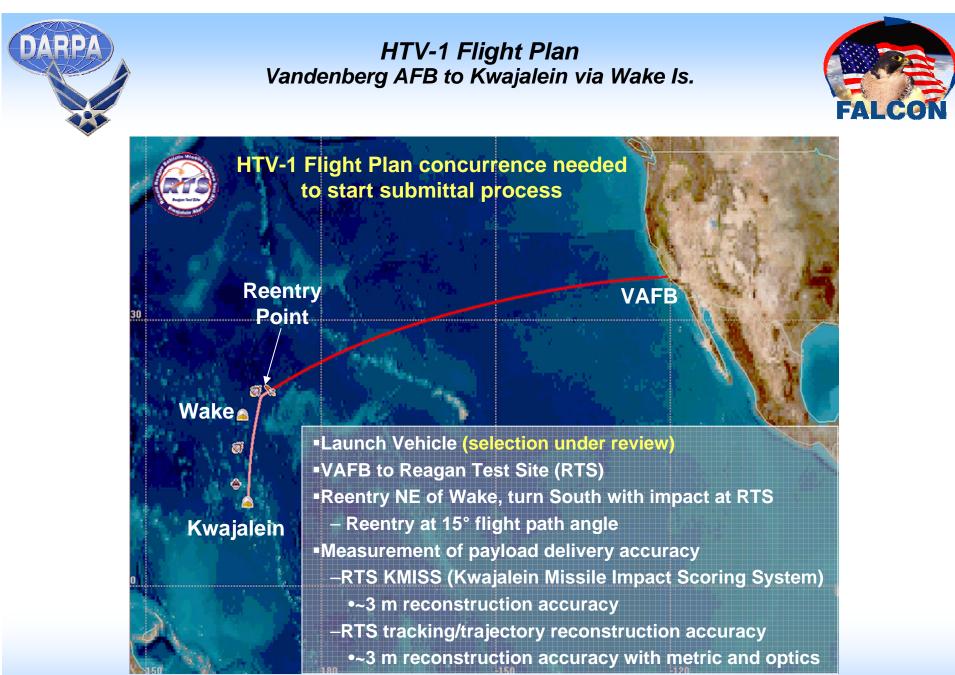


Antenna

ABRO VENICLE

> Government **Furnished Booster**

Launched





# HTV-2 Objectives



East launch, 28.5° latitude, VE = 23,500 ft/sec, hE = 350 kft

Key Objectives

- ➢Payload ~ 1000 lbs (TBD)
- ➤Gross weight = 2000 lbs
- Downrange = 9000 nm
- Crossrange = 3000 nm
- Accuracy = 3 m CEP
- ≻Global coverage
- Recallable and re-targeting capability

Additional Objectives

- Impact velocity ~4 kfps (TBD)
- All-azimuth terminal maneuver capability
- Carriage & high-speed dispensing of payloads (TBD)
- Minimize collateral damage

#### HTV-2 meets all objectives to provide flexible global capabilities



# HTV-2 Demonstration System Summary

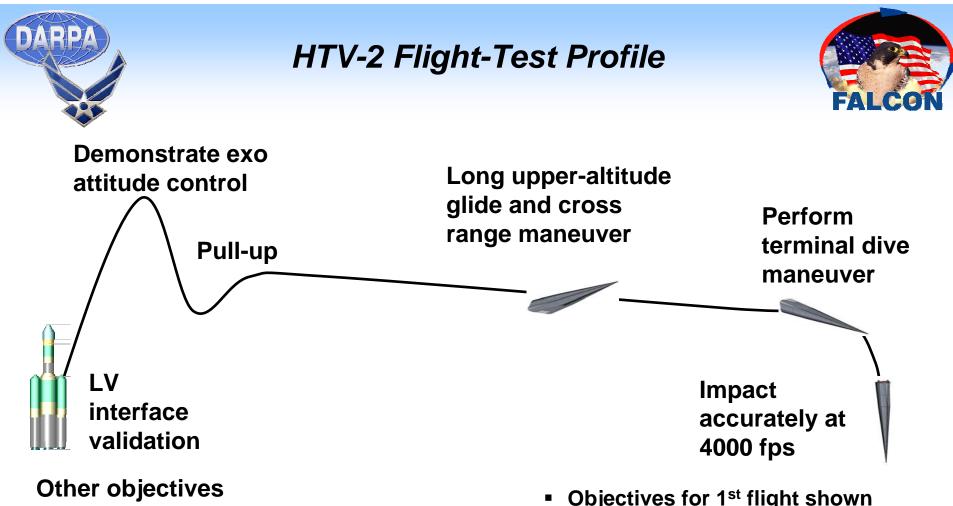


- Thermal protection
  - Low recession carbon-carbon aeroshell
  - Advanced Multi-Layer Insulation for long duration reentry flight
- Aerodynamic performance
  - Extended range through high L/D
  - Sharp Leading Edge Design
- NG&C performance
  - Significant maneuverability required for terminal impact

HTV-2

- Communications
  - Maintain up/downlink throughout long-range flight

HTV-2 Demonstrates Enabling Hypersonic Technologies for future Common Aero Vehicle Operational System



- Maintain flight safety throughout
- GPS acquisition during boost phase
- Command/telemetry link throughout mission
- Objectives for 1<sup>st</sup> flight shown
- 2<sup>nd</sup> flight options
  - repeat 1<sup>st</sup> with equipment updates
- perform payload dispensing demonstration

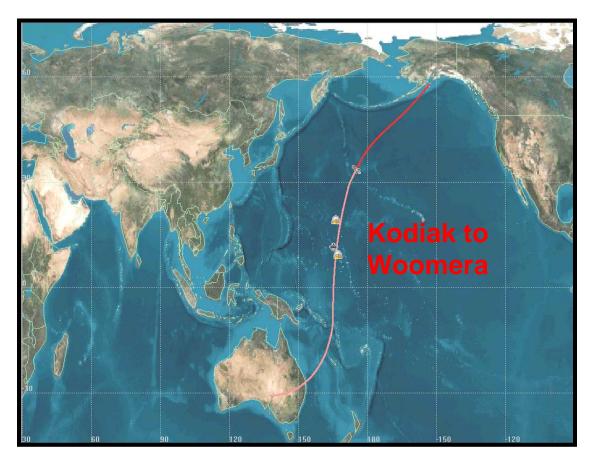
#### Our flight profile demonstrates all important performance attributes



# **Proposed HTV-2 Flight Plan**



- SLV launch to reentry at 23,000 fps and 5° path angle
- Kodiak to Woomera (6200 nm) via Wake and Kwajalein
- Continuous tracking and telemetry use extended mobile range, AFSCN and MILSATCOM assets
- Overflight of populated Australian east coast
  - Flight crosses coast >100 kft altitude
  - Controlled Flight Termination available
- Terminal phase tracking available from Woomera



#### HTV-2 flight plan provides long-duration test environments



# ECAV-OS Weight Traceability



	Weight (Ibm)			Reduction Plan	
			OS	Reduction Flan	
Payload			960	Customer Provided	
Heatshield			400	Analysis refinement	
Insulation			90	Analysis refinement	
Structure			240	Analysis – g load reduction	
Electronics			85	Actuator/battery refinement	
Ballast			225	Structure/HS reduction, CG movement/aero refinement	
Total			2000		

Roadmap defined to achieve 2000 lb ECAV-OS mass

Ref # 5.1.1 b

# **HTV-3 Demonstration System Summary**

HTV-3



- Thermal Protection System (TPS)

   TPS/Structure demonstrate capability for HCV-OS environment
- Aerodynamic performance
  - Shaped to demonstrate high aerodynamics, aerothermodynamics and flight control of HCV-OS
- Maximize Reuse
  - Multiple flight tests demonstrate system reusability/TPS refurbishment
- Builds upon HTV-1 and HTV-2 technologies

   Technology risks minimized while payoff is maximized

HTV-3 Demonstrates Enabling Hypersonic Technologies for future Hypersonic Cruise Vehicle Operational System



HCV-OS

#### HTV-3 Design & Capabilities



#### HCV-OS primary objectives trace to HTV-DS design capabilities

Mission Requirement	OS Design Objective	DS Verification Compliance		
Global Reach 9000nm in 2 hours	Mission performance via high L/D osculating flowfield waverider configuration	Shaped to demonstrate high L/D wave-rider osculating flowfield aerodynamics, aerothermodynamics, flight control		
	High tolerance to thermal environment	Demonstrate enabling TPS / structural technologies in OS flight environment		
Aircraft-like operations	Reusability with rapid 12 hr turnaround and minimum maintenance	Perform multiple flight tests, demonstrating system reusability / TPS refurbishment		

#### Demonstrate Key Enabling Technologies

Osculating Flowfield Waverider shape 4000°F class passive TPS 3000°F class passive TPS Lightweight acreage passive TPS 'Warm' structure-tankage-TPS integration



Low Risk Demo

#### HTV-3 is directly traceable to HCV-OS to mitigate risk of key enabling technologies



# FALCON Materials IPT



# Materials IPT (MIPT) focusing on materials issues (TPS and hot structures) for HTV-2 and HTV-3

- HTV-1 is assumed to utilize state-of-the-art materials
- In the initial phases of the MIPT, only Airframe technology is considered. Propulsion hot structures will be considered at a later date as required/requested.

# >MIPT objectives

- Evaluate relevancy of ongoing government funded materials/design efforts to HTV-1, HTV-2, HTV-3, ECAV-OS, and HCV-OS needs.
- Work with the FALCON prime to develop a materials plan that integrates MIPT efforts with contractor efforts
- Recommend supplemental and new start efforts to fill technology gaps

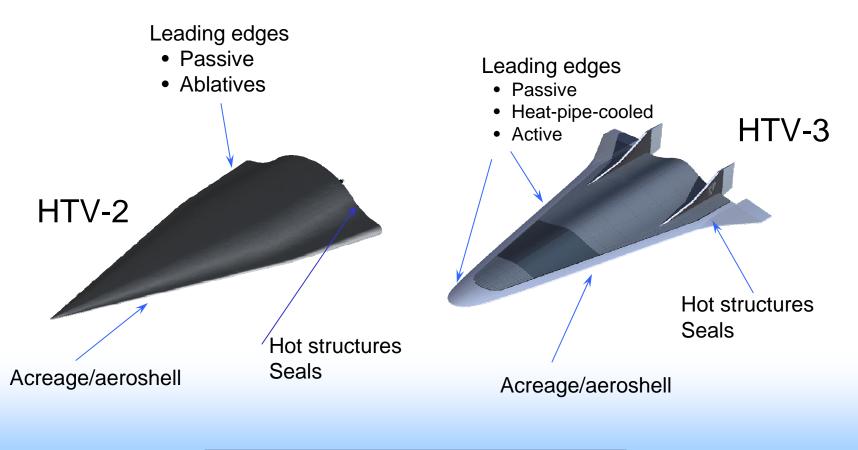


#### Critical Airframe Components TPS/Hot Structures



# Enhanced CAV (OS)

# Hypersonic Cruise Vehicle (OS)





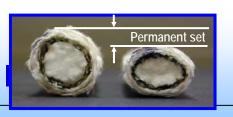
- C-CAT
- Pratt & Whitney

## >3600°F Refractory **Composites**

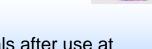
- Physical Sciences, Inc. (PSI)
- Composite Innovations, Inc. (CIC)
- ATK

# High Temperature Multi-**Layer Insulation**

 Refractory Composites, (RCI)



#### Seals after use at 1900°F





Sharp leading edge,  $T > 3600^{\circ}F$ 

FEA of TPS attachment





German multi-layer insulation

surface

MIPT FY04 Activitie

Contracted Through UDR

X-37 C/C control



FALCON... Enabling future hypersonic technologies



Unprecedented hypersonic technology validation in flight

Building block approach maximizes payoff while minimizing technology risks

TPS is the key technology

<u>Newly Established STRATCOM October, 2002</u> Unified Command - Given Global Strike Mission: "Establish and provide full-spectrum global strike... to meet both deterrent and decisive national security objectives" "The capability to plan for and deliver rapid, limited-duration, extended-range precision kinetic and non-kinetic effects"

FALCON will demonstrate technologies required for tomorrow's global reach missions!