

ARMY RESEARCH LABORATORY



Thermal Degradation Effects on Consolidation and Bonding in the Thermoplastic Fiber-Placement Process

by Bruce K. Fink, John W. Gillespie, Jr., and Nuri B. Ersoy

ARL-TR-2238

June 2000

DEC QUALITY INSPECTED 4

Approved for public release; distribution is unlimited.

20000724 038

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

Citation of manufacturer's or trade names does not constitute an official endorsement or approval of the use thereof.

Destroy this report when it is no longer needed. Do not return it to the originator.

Abstract

Effects of elevated temperature exposure during thermoplastic fiber placement on bonding and consolidation are investigated experimentally for AS4/polyetherketoneketone (PEKK) composite. Coupons of 24 layers are consolidated on the University of Delaware Center for Composite Materials (UD-CCM) fiber-placement robot at deposition rates of 20 and 40 mm/s over a range of process temperatures (700–900 °C, with 50 °C increments). The main torch and preheater distances and the compaction force are held constant for all coupons. Two competing mechanisms governing strength buildup are considered: (1) polymer bonding and (2) degradation. Coupons are sectioned, and one-half of each coupon is reconsolidated in a hot press at conventional processing conditions (i.e., 30 min at 370 °C and 0.70-MPa pressure) in order to remove any effect of poor consolidation on strength. Void content of the robot-consolidated panels is measured. Short-beam shear (SBS) tests are performed on the specimens cut from each coupon. Strength and void-content measurements for robot-consolidated panels are presented to illustrate the effect of processing parameters on product quality. Results of the SBS strength tests performed on reconsolidated coupons indicate that there is significant decrease in the strength of coupons consolidated at 20-mm/s deposition rate and high torch temperatures, possibly due to polymer degradation, whereas approximately the same value of reconsolidated strength is measured for the 40-mm/s deposition rate, suggesting that polymer degradation is insignificant at that rate. Effects of void content on SBS tests are discussed.

Table of Contents

	<u>Page</u>
List of Figures	v
List of Tables	vii
1. Introduction	1
2. Experimental Work	1
2.1 Fiber-Placement Equipment.....	1
2.2 Materials.....	2
2.3 Preparation of Test Coupons.....	3
2.4 SBS Tests.....	4
2.5 Microscopic Examinations.....	5
3. Results and Discussion	6
3.1 SBS Tests.....	6
3.2 Microscopic Examinations.....	9
3.3 Effect of Voids on SBS Strength.....	10
4. Conclusions	17
5. References	21
Appendix A: Thermal Gravimetric Analysis of Degradation of Polyetherketoneketone (PEKK)	23
Appendix B: Fourier Transform Infrared Spectroscopic Study of Thermal Degradation of AS4/Polyetherketoneketone (PEKK) Composite	31
Distribution List	37
Report Documentation Page	57

INTENTIONALLY LEFT BLANK.

List of Figures

<u>Figure</u>	<u>Page</u>
1. Schematic of Fiber-Placement Process Showing Components.....	3
2. Schematic of Fiber-Placement Process Showing Variables.....	4
3. Schematic of SBS Test.....	6
4. Effect of Torch Temperature on SBS Strength for 20-mm/s Deposition Rate.....	8
5. Effect of Torch Temperature on SBS Strength for 40-mm/s Deposition Rate.....	8
6. Effect of Processing Conditions on Void Content	10
7. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 700 °C Torch Temperatures and 20-mm/s Deposition Rate	11
8. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 850 °C Torch Temperatures and 20-mm/s Deposition Rate	12
9. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 700 °C Torch Temperatures and 40-mm/s Deposition Rate	13
10. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 850 °C Torch Temperatures and 40-mm/s Deposition Rate	14
11. Effect of Void Content on SBS Strength for 20-mm/s Deposition Rate.....	16
12. Effect of Void Content on SBS Strength for 40-mm/s Deposition Rate.....	17
13. Micrograph of a Typical SBS Specimen Consolidated at 700 °C Torch Temperatures and 20-mm/s Deposition Rate	18
14. Micrograph of a Typical SBS Specimen Consolidated at 850 °C Torch Temperatures and 20-mm/s Deposition Rate	18
A-1. TGA Diagram of PEKK Neat Films	25
A-2. TGA Diagram of AS4/PEKK Composite	26
A-3. Infrared Camera Readings of Surface Temperature at 20-mm/s Deposition Rate	27

<u>Figure</u>	<u>Page</u>
A-4. Model Predictions for Surface Temperature at 20-mm/s Deposition Rate	28
A-5. Maximum Surface Temperature for Various Torch Temperature Settings at 20-mm/s Deposition Rate.....	29
B-1. Diffuse Reflectance Spectra for 700 °C Torch Temperature	34
B-2. Diffuse Reflectance Spectra for 800 °C Torch Temperature	35
B-3. Diffuse Reflectance Spectra for 900 °C Torch Temperature	36

List of Tables

<u>Table</u>		<u>Page</u>
1.	Process Variables and Settings.....	4
2.	SBS Data for 20-mm/s Deposition Velocity.....	7
3.	SBS Data for 40-mm/s Deposition Velocity.....	7
4.	Void Content Measurements for 20-mm/s Deposition Velocity.....	9
5.	Void Content Measurements for 40-mm/s Deposition Velocity.....	9
6.	Normalized Strength Data for 20-mm/s Deposition Velocity.....	15
7.	Normalized Strength Data for 40-mm/s Deposition Velocity.....	16
A-1.	Maximum Surface Temperatures for Various Torch Temperature Settings at 20-mm/s Deposition Rate.....	28

INTENTIONALLY LEFT BLANK.

1. Introduction

The recent advances in automated manufacturing technology of composite materials resulted in a highly promising process called fiber placement. In the fiber-placement process, as opposed to filament winding, (1) the fibers are not restricted to near-geodesic paths, (2) paths can start and end at any point, (3) the fibers can be placed along concave mandrel sections, and (4) the amount of fiber and the width of the fiber band can be changed along the width of the delivery path. This makes it possible to fabricate very general nonaxisymmetric shapes with varying thickness.

The advantages of fiber-placement processes can be combined with the high throughputs that can be achieved using thermoplastic prepreg tape that can be shaped and consolidated on line. This is due to the chemical structure of the resin that enables melting under heat and consolidation before cooling. On-line consolidation technology thus eliminates the use of an autoclave by applying heat directly at the nip-point, welding the tape on the preceding layer under compaction pressure, and consolidating shortly behind the nip-point where the temperature is lower.

Mechanisms governing strength buildup and consolidation during thermoplastic fiber-placement processes should be well understood in order to control the processing parameters to obtain the optimum product quality.

2. Experimental Work

2.1 Fiber-Placement Equipment. At the University of Delaware Center for Composite Materials (UD-CCM), the first experiences with thermoplastic filament winding and tape placement were gained by Wells and Steiner [1]. Considerable improvements to the entire robotic fiber-placement workcell concerning heating sources, mandrel manipulator, placement head capabilities, and process control have been integrated as a result of several investigations [2, 3].

The centerpiece of the robotic tape-placement workcell used in this investigation was a Model 762 AEG-Westinghouse PUMA industrial robot. It has a repeatability of 0.02 mm, six axes, a maximum reach of 1,244 mm, and a payload of 200 N. The PUMA control system operates on the VAL II high-level language. It can be linked to the environment through 32 input/output ports and control signals.

The prototype tape-placement head is mounted to the robot arm by a pair of linear bearings. As shown schematically in Figure 1, thermoplastic prepreg tape is fed from a supply spool and guided through several eyelets, the rollers of the tape start lever, and the cutting unit, which allows cutting and refeeding of the tape. The magnetic brake attached to the supply spool exerts a constant tape tension. After passing the cutter, the incoming tape, as well as the formerly placed substrate, are preheated by the first hot gas torch. The tape is then guided through the lay-down roller that establishes contact between the substrate and the preheated tape. The second torch heats through the thickness to melt the entire tape and tape-substrate interface. Passing the air-cooled compaction roller, the tape consolidates onto the substrate under pressure and cools down, while the voids within and between the tape and substrate are reduced and bonding develops. In this way, the tape can be placed on a flat mandrel by laying one strip next to another.

The hot gas torches are fed with nitrogen. The flow rate and the temperature of the outcoming gas for two torches are controlled separately by a hot gas torch control system manufactured by Automated Dynamics Corporation. National Instruments LabVIEW-based process control software is linked to the robot controller and the torch controller, which allows for on-line setting of the process parameters such as gas flow rate, torch temperatures, and compaction force.

2.2 Materials. The material used for experiments was AS4 graphite/polyetherketoneketone (PEKK) with 58% fiber-volume fraction. The void content of the incoming tow is measured to be $3.8 \pm 0.8\%$. PEKK was developed by DuPont as a high-performance thermoplastic matrix system for advanced composites. The carbon-fiber-reinforced composite laminates with

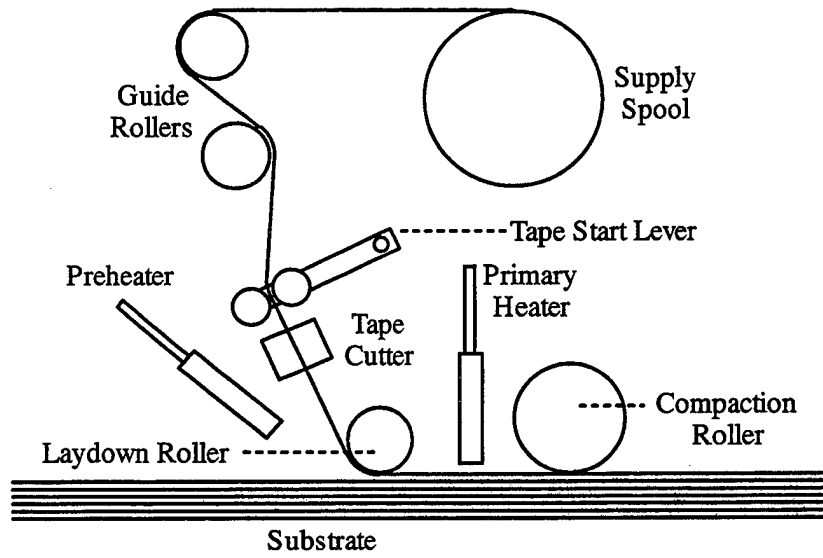


Figure 1. Schematic of Fiber-Replacement Process Showing Components.

PEKK matrix, prepared from the propriety melt impregnated tows, show high flexural, shear, and compressive strengths and good environmental durability and hot-wet stability.

2.3 Preparation of Test Coupons. Test coupons were consolidated on a graphite plate with deposition rates of 20 and 40 mm/s. Both torch temperatures were varied from 700 to 900 °C with 50 °C increments. The main torch and preheater distances and the compaction force were held constant for all test coupons. Figure 2 shows the process parameters, and Table 1 shows the settings used for consolidating the test coupons. For the case of 900 °C torch temperatures and 20-mm/s deposition rate, a test coupon could not be consolidated because excessive polymer degradation prevents deposition of fresh tows. After consolidating on the robot, test coupons were sectioned, and one-half of each coupon was reconsolidated in the hot press at conventional processing conditions (i.e., 30 min at 370 °C under 0.70-MPa pressure) in order to remove any effect of poor consolidation and bonding on strength. Later, short-beam shear (SBS) specimens were cut from both the robot-consolidated and reconsolidated coupons.

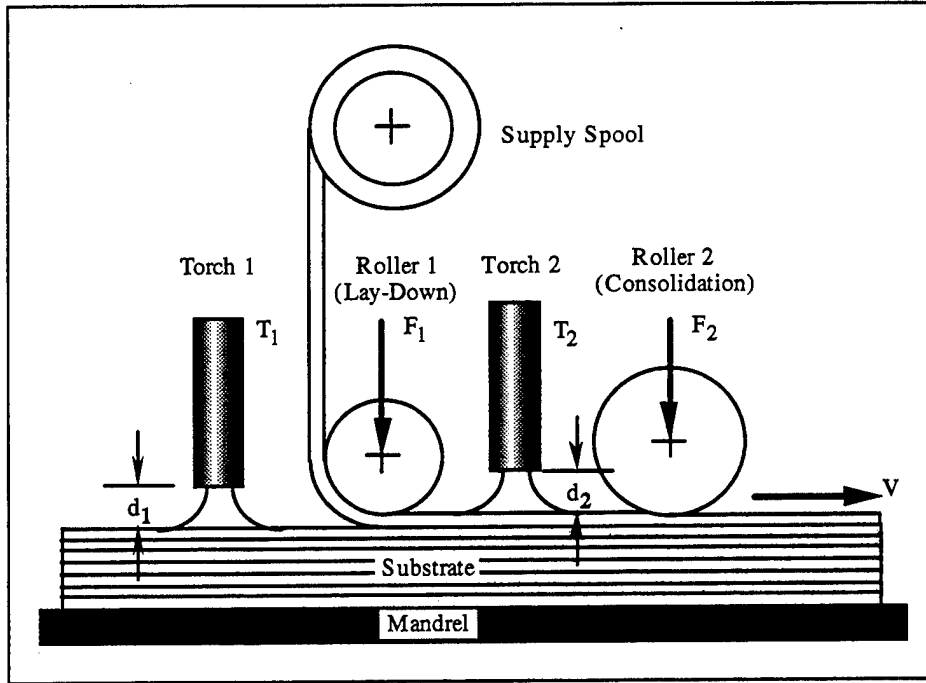


Figure 2. Schematic of Fiber-Placement Process Showing Variables.

Table 1. Process Variables and Settings

Process Variable	Abbreviation	Settings
Deposition Rate	V	20 and 40 mm/s
Compaction Force ^a	F ₂	320 N
Preheater Temperature	T ₁	700–900 °C, with 50 °C increments
Preheater Distance ^a	D ₁	20 mm
Main Torch Temperature	T ₂	700–900 °C, with 50 °C increments
Main Torch Distance ^a	D ₂	4 mm

^a Real-time computer-controlled process variables.

2.4 SBS Tests. The SBS strength of test coupons was measured according to the American Society for Testing of Materials (ASTM) D 2344-84 [3, 4]. Accordingly, this method is useful for quality control and specification purposes and is also applicable for research and development purposes concerned with interply strength. Although the apparent shear strength obtained in this test cannot be used as a design criterion, it can be utilized for comparative testing of composite materials, if all failures are in horizontal shear.

SBS specimens were cut from test coupons by a diamond saw to nominal dimensions of 6.35 mm width and six times the specimen thickness. Although 24 layers were deposited for each test coupon, the thickness of the coupons varied with processing parameters; hence, the thickness of the specimens was an uncontrolled parameter. Ten specimens were used for each case in order to obtain a satisfactory average. The thickness and width of each specimen were measured to the nearest 0.025 mm at midpoint. The specimens were placed in the test fixture, as shown in Figure 3. The side supports of the fixture were pushed to the span of four times the specimen thickness. The specimens were aligned so that the midpoint was centered and the long axis was perpendicular to the cylindrical axis (under the loading nose). The tests were performed using an Instron 1125 screw-driven test frame at room temperature using 454-kg (1,000 lb) load cell at full scale and 1.3-mm/min (0.05 in/min) crosshead speed. The failure load was recorded as the first maximum load in the load-displacement diagram. The shear strength was calculated as

$$S_H = 0.75 \frac{P_B}{bd}, \quad (1)$$

where S_H is the shear strength, P_B is the breaking load, b is the width, and d is the thickness of specimen. Arithmetic mean and standard deviation of values obtained for each test coupon were calculated.

2.5 Microscopic Examinations. Samples for microscopic examination were prepared by cutting the robot-consolidated test coupons by a diamond saw perpendicular to the fiber direction. Samples were embedded in epoxy resin and ground using an automated fixture on 180-, 360-, and 600-grid silicon carbide paper and then finished on 12.5-, 9.5-, 5-, 3-, and 1- μ m polishing cloths, applying the specified alumina powder/water suspension. An ultrasonic water bath was used to clean the samples between successive steps. The samples were examined under reflected light with 100 \times magnification. Ten snapshots of capturing frames at the surface and middle of each test coupon were recorded to videotape and imported to a Macintosh Quadra 650 workstation equipped with a video-capture board. Void-content measurements were performed with NIH Image 1.47 image analysis software.

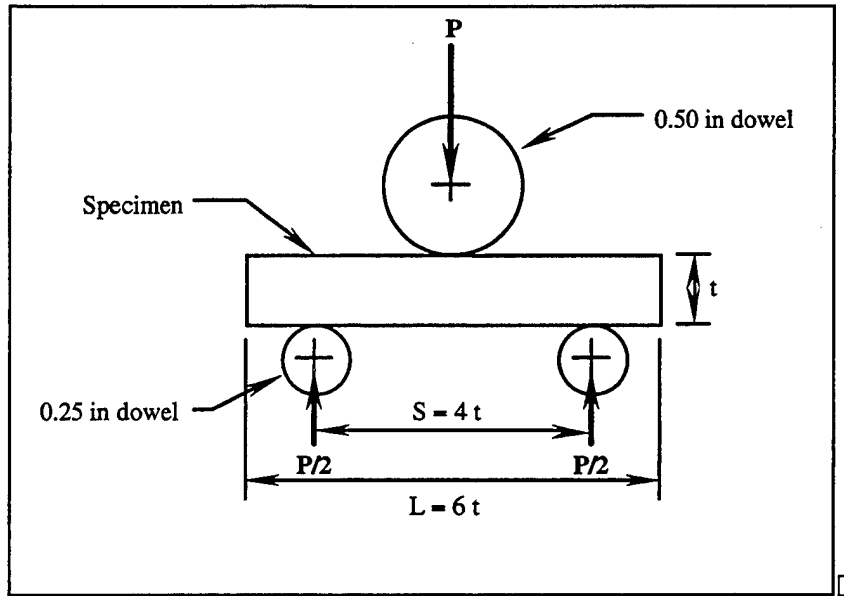


Figure 3. Schematic of SBS Test.

3. Results and Discussion

The strength and void-content measurement results from the fiber-placement process are presented with a two-fold objective: (1) to illustrate the influence of the physical processing parameters on the bond-strength development and void-content reduction and (2) to delineate the relative effects of two competing mechanisms governing strength buildup (polymer bonding and degradation).

3.1 SBS Tests. Arithmetic mean and standard deviation of values obtained from 10 SBS samples for each test coupon were calculated and are tabulated in Tables 2 and 3 for 20- and 40-mm/s deposition rates, respectively. From Table 3, for 40-mm/s deposition rate, approximately the same value strength is obtained upon reconsolidation. Hence, the average of the reconsolidated strength values for that deposition rate, which comes out to be 82.6 ± 2.9 MPa (11.98 ± 0.42 Ksi), is used as a baseline.

In Figures 4 and 5, SBS strength values of the robot-consolidated and reconsolidated coupons are plotted vs. torch temperature for 20- and 40-mm/s deposition rates, respectively. As can be

Table 2. SBS Data for 20-mm/s Deposition Velocity

Torch Temperature (degrees Celsius)	Robot-Consolidated SBS Strength (MPa [Ksi])	Reconsolidated SBS Strength (MPa [Ksi])
700	44.9 ± 6.1 [6.51 ± 0.89]	80.2 ± 2.2 [11.63 ± 0.32]
750	56.5 ± 2.8 [8.19 ± 0.41]	80.6 ± 3.2 [11.69 ± 0.47]
800	65.1 ± 2.5 [9.44 ± 0.36]	73.4 ± 2.3 [10.64 ± 0.34]
850	67.2 ± 1.3 [9.75 ± 0.19]	59.0 ± 2.6 [8.55 ± 0.38]

Table 3. SBS Data for 40-mm/s Deposition Velocity

Torch Temperature (degrees Celsius)	Robot-Consolidated SBS Strength (MPa [Ksi])	Reconsolidated SBS Strength (MPa [Ksi])
700	43.4 ± 5.3 [6.30 ± 0.77]	79.4 ± 1.5 [11.52 ± 0.22]
750	49.9 ± 6.0 [7.23 ± 0.87]	81.4 ± 3.3 [11.80 ± 0.48]
800	55.8 ± 4.9 [8.09 ± 0.71]	82.9 ± 2.8 [12.02 ± 0.41]
850	64.1 ± 2.2 [9.30 ± 0.32]	83.6 ± 4.7 [12.12 ± 0.68]
900	62.5 ± 4.3 [9.07 ± 0.63]	82.8 ± 2.9 [12.00 ± 0.42]

seen from Figure 4, there is a considerable drop in reconsolidated strength for torch temperatures of 800 and 850 °C, indicating that full strength represented by the baseline cannot be recoverable upon reconsolidation. At the deposition rate of 40 mm/s, the dwell time of the polymer at high temperatures under preheat and main torches is shorter, and polymer thermal degradation is expected to be insignificant and to not affect further strength buildup upon reconsolidation.

The origin of polymer degradation is uncertain. It may be due to (1) decomposition of the polymer matrix at the surface of the tow due to exposure of high-temperature stream of nitrogen gas, reducing the thickness of the resin rich layer and hence unveiling the carbon fibers, or (2) cross-linking or cyclization of the polymer chains, reducing the chain mobility and hence, ability to diffuse and bond. Fourier-transform infrared (FTIR) and thermal gravimetric analysis (TGA) characterization studies of neat PEKK films and PEKK-based composites are presented in the Appendices. While degradation resulting in weight loss can be monitored with these techniques, the temperatures at which weight-loss degradation occurs is presumed to be

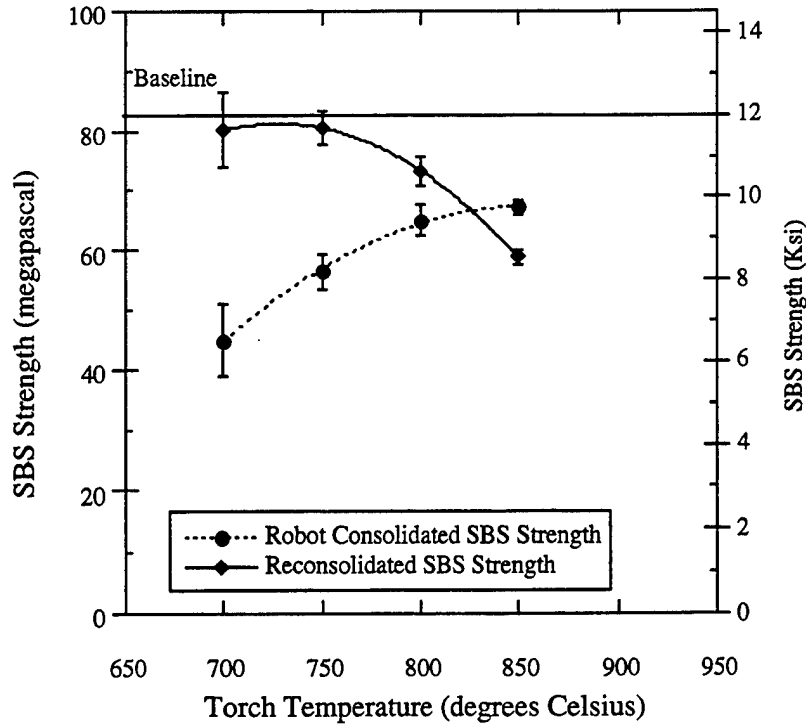


Figure 4. Effect of Torch Temperature on SBS Strength for 20-mm/s Deposition Rate.

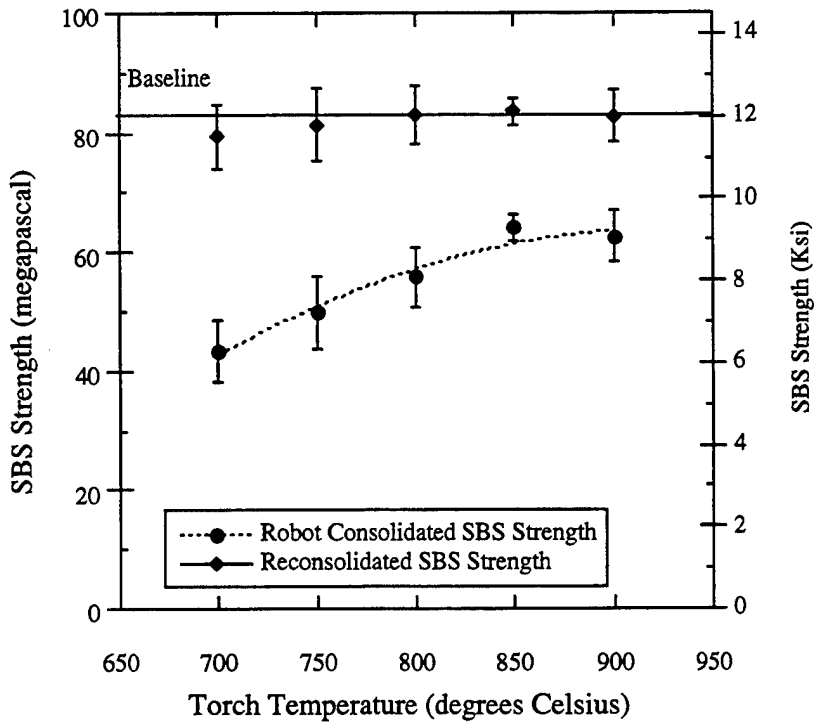


Figure 5. Effect of Torch Temperature on SBS Strength for 40-mm/s Deposition Rate.

purposefully avoided in the manufacturing process. The origin of nonweight-loss polymer degradation and its effect on bonding needs further investigation.

3.2 Microscopic Examinations. Arithmetic mean and standard deviation of void content values in the bulk of each test coupon are calculated and tabulated in Tables 4 and 5 for 20- and 40-mm/s deposition rates, respectively. In Figure 6, void contents of the test coupons are plotted against torch temperature. For high temperatures, the difference between the surface and bulk void contents is higher. At these temperatures, the region at the melt state is expected to be deeper, and void reduction and further consolidation are expected to take place upon subsequent passes when fresh tows are placed on already consolidated substrates.

Table 4. Void Content Measurements for 20-mm/s Deposition Velocity

Torch Temperature (degrees Celsius)	Void Content (bulk) (percent)	Void Content (surface) (percent)
700	1.60 ± 0.48	4.00 ± 0.80
750	1.27 ± 0.25	2.48 ± 0.38
800	1.13 ± 0.23	2.37 ± 0.30
850	1.00 ± 0.19	3.06 ± 0.67

Table 5. Void Content Measurements for 40-mm/s Deposition Velocity

Torch Temperature (degrees Celsius)	Void Content (bulk) (percent)	Void Content (surface) (percent)
700	2.58 ± 0.95	3.27 ± 0.87
750	1.55 ± 0.37	2.16 ± 0.41
800	0.84 ± 0.24	1.07 ± 0.19
850	0.37 ± 0.09	1.27 ± 0.40
900	0.33 ± 0.09	1.66 ± 0.55

Figures 7–10 show typical micrographs at the surface and in the bulk of the cross sections perpendicular to the fiber direction for four extreme cases. As can be seen qualitatively from these micrographs, most of the voids are located at the intralaminar region, and interlaminar

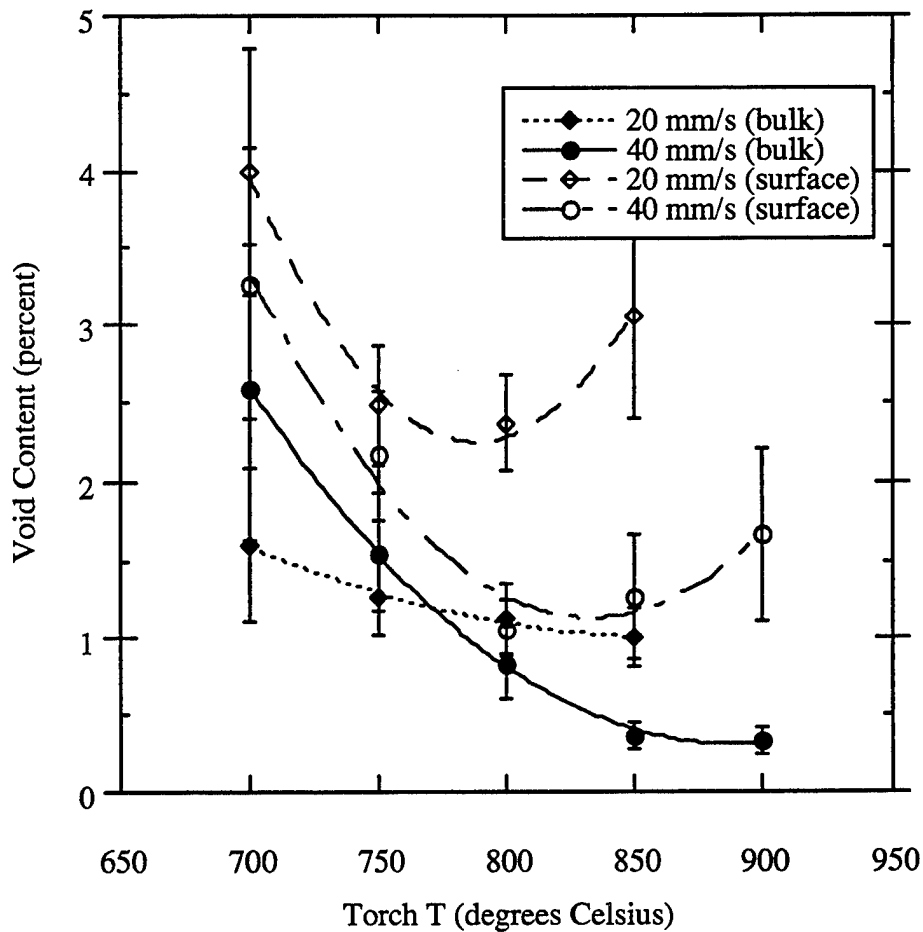
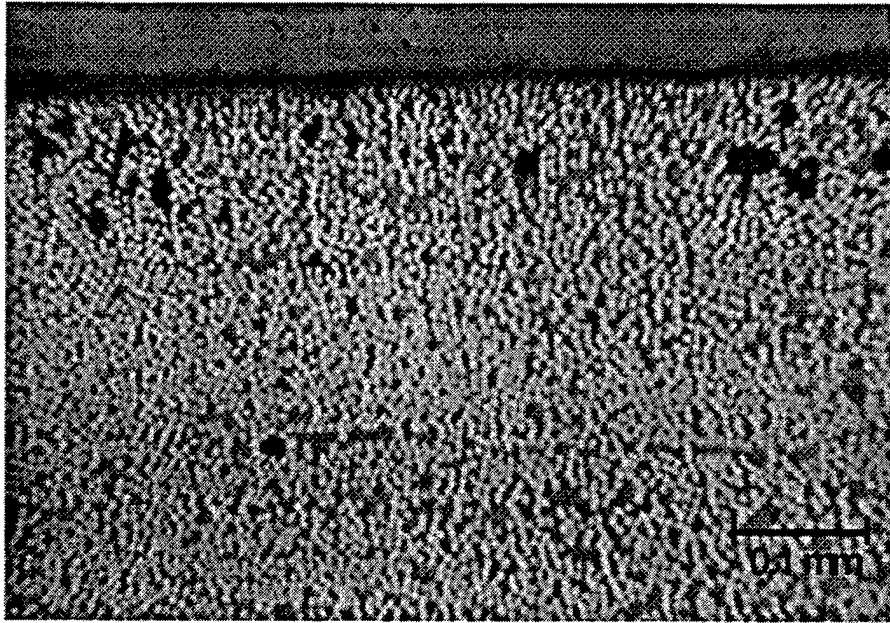


Figure 6. Effect of Processing Conditions on Void Content.

voids and regions of poor bonding can be seen for low torch temperatures at both deposition rates (Figures 7 and 9). Bigger voids in the topmost layer are evidence of void growth promoted by high torch temperatures and long dwell times.

3.3 Effect of Voids on SBS Strength. As indicated in the previous section, most of the voids observed were located in the intralaminar region. These voids may generate stress concentrations and may promote intralaminar fracture during short-beam shear tests, complicating the task of obtaining true interlaminar bond strength. Hence, a study was conducted to evaluate the effect of voids on the measured interlaminar shear strength.

(a)



(b)

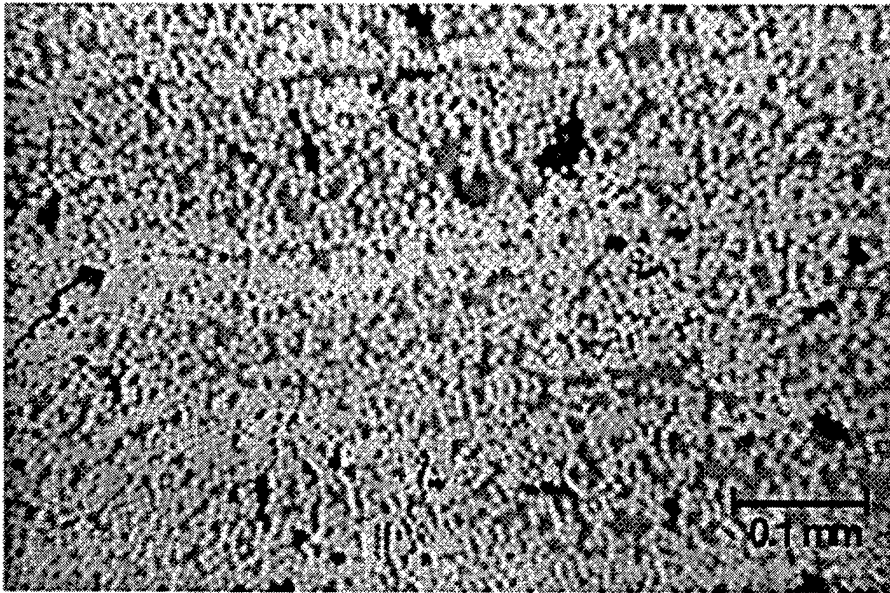
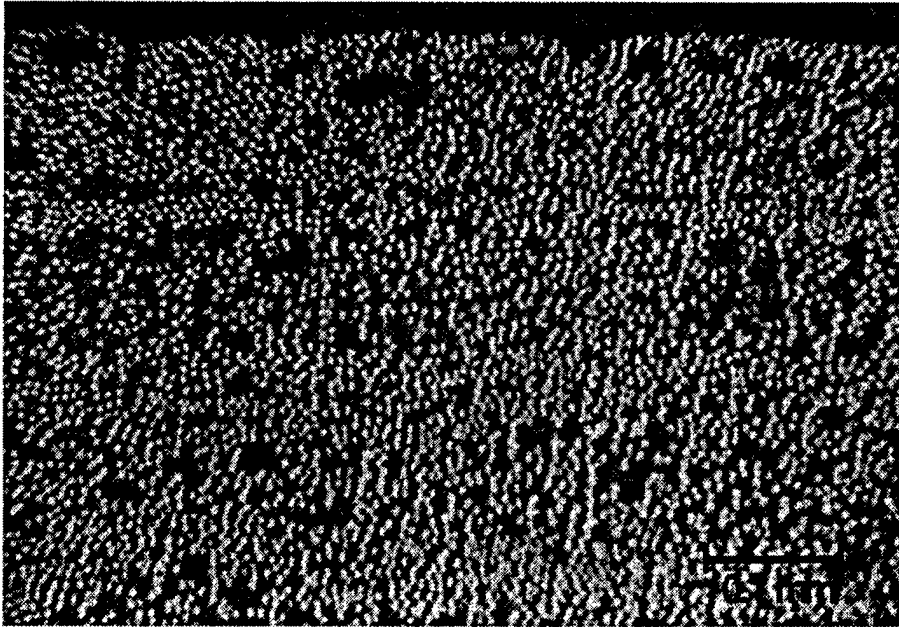


Figure 7. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 700 °C Torch Temperatures and 20-mm/s Deposition Rate.

(a)



(b)

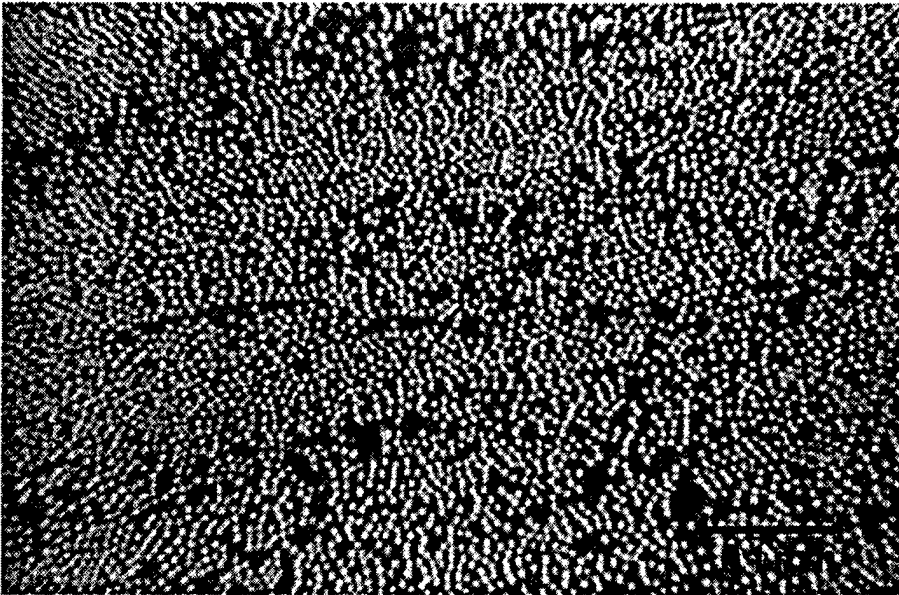
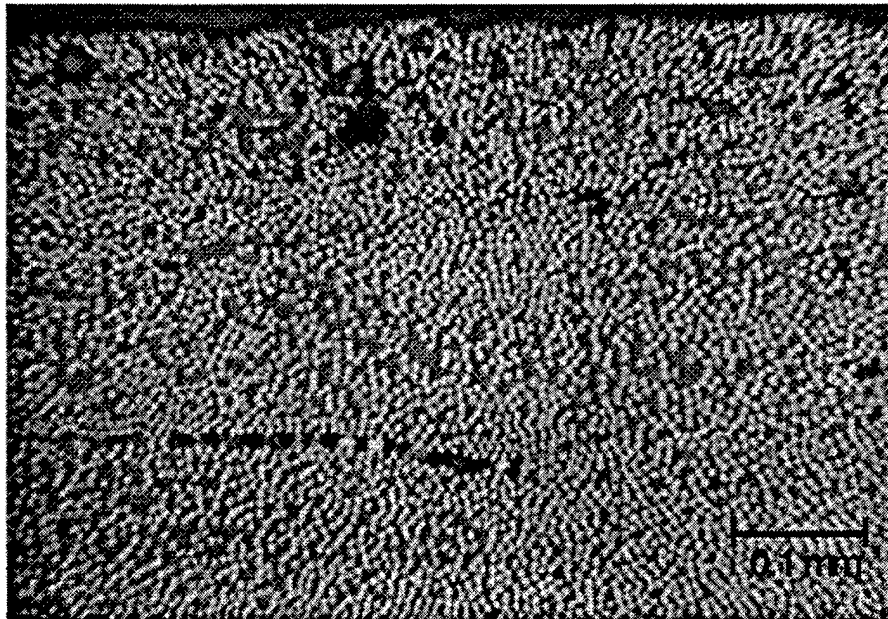


Figure 8. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 850 °C Torch Temperatures and 20-mm/s Deposition Rate.

(a)



(b)

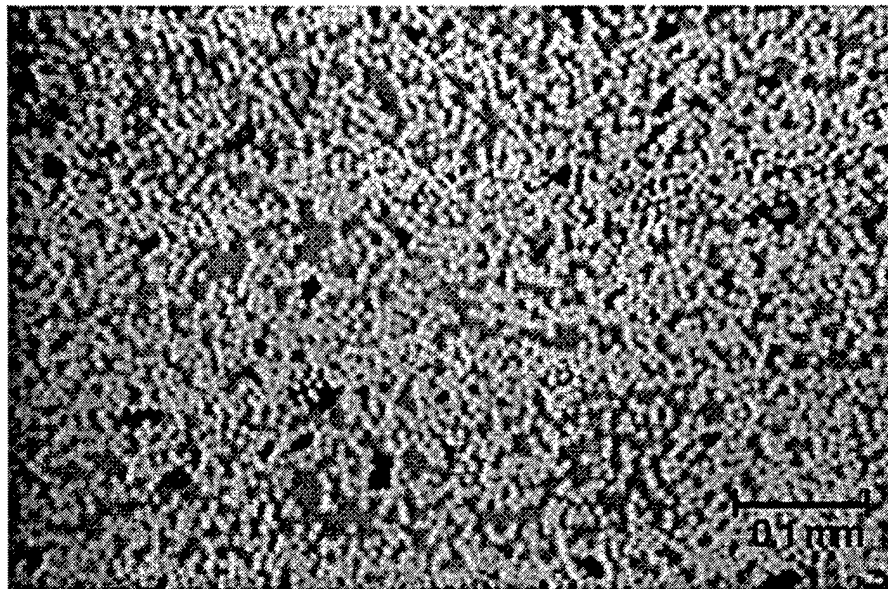
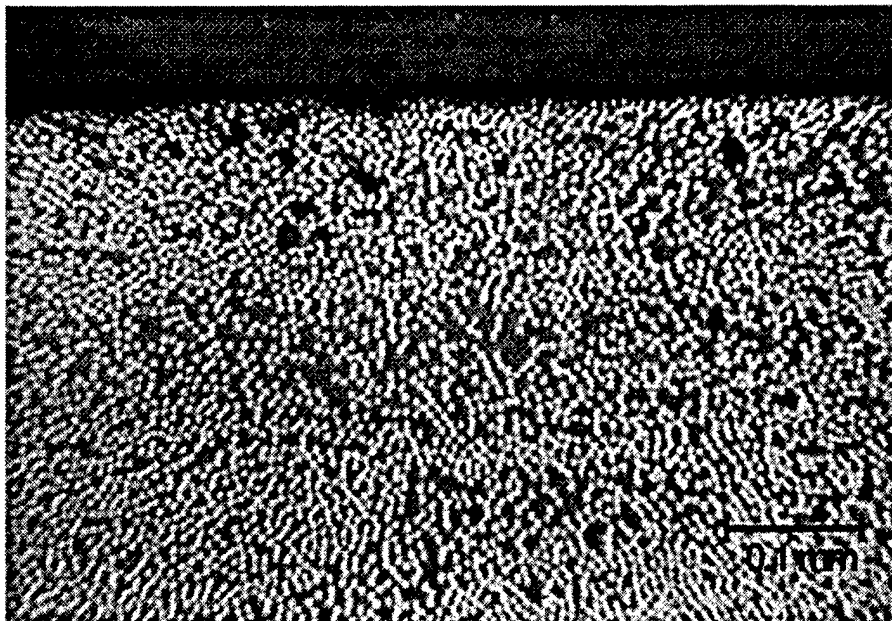


Figure 9. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 700 °C Torch Temperatures and 40-mm/s Deposition Rate.

(a)



(b)

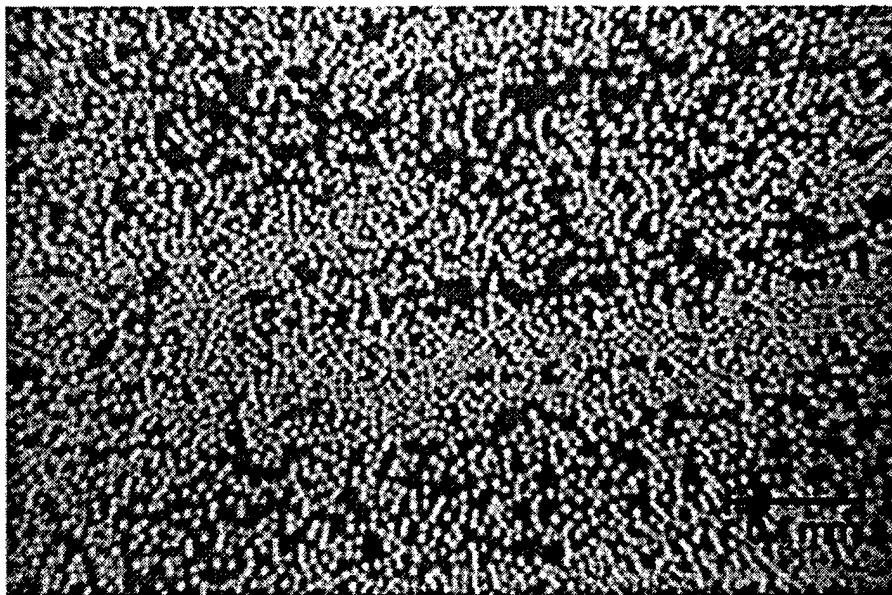


Figure 10. Typical Micrographs of the (a) Surface and (b) Bulk of the Test Coupon Consolidated at 850 °C Torch Temperatures and 40-mm/s Deposition Rate.

The effect of void fraction on interlaminar shear strength has been analyzed by Greszczuk [5], Bowles and Frimpong [6], and by Olson et al. [7]. Bowles and Frimpong theoretically derived the following equations for two possible configurations for voids in composites—cylindrical and spherical:

cylindrical:
$$ILSS_r = 1 - [4V_v / 3.14(1 - V_{fv})]^{1/2}, \quad (2)$$

spherical:
$$ILSS_r = 1 - 0.785[6V_v / 3.14(1 - V_{fv})]^{2/3}, \quad (3)$$

where V_{fc} is the fiber-volume fraction of the composite with voids, V_v is the void-volume fraction, and $ILSS_r$ is the interlaminar shear strength of the composite with voids relative to that of the void-free composite.

In Tables 6 and 7, the $ILSS$ values calculated using equations (2) and (3) for both cylindrical and spherical voids, together with normalized SBS strength data, are tabulated for 20- and 40-mm/s deposition rates, respectively. A baseline strength value of 82.6 MPa is used to normalize the strength data. In Figures 11 and 12, these values are plotted against the torch temperature. $ILSS_r$ calculated for cylindrical voids yields a more conservative estimate, and most of the voids observed in robot-consolidated panels were of cylindrical geometry.

Table 6. Normalized Strength Data for 20-mm/s Deposition Velocity

Torch Temperature (degrees Celsius)	Robot-Consolidated SBS Strength	Reconsolidated SBS Strength	$ILSS_r$ (Sph. Voids)	$ILSS_r$ (Cyl. Voids)
700	0.55 ± 0.07	0.98 ± 0.03	0.78	0.86
750	0.69 ± 0.03	0.98 ± 0.04	0.80	0.88
800	0.79 ± 0.03	0.89 ± 0.03	0.81	0.89
850	0.82 ± 0.02	0.72 ± 0.03	0.83	0.90

Table 7. Normalized Strength Data for 40-mm/s Deposition Velocity

Torch Temperature (degrees Celsius)	Robot-Consolidated SBS Strength	Reconsolidated SBS Strength	ILSS _r (Sph. Voids)	ILSS _r (Cyl. Voids)
700	0.55 ± 0.07	0.98 ± 0.03	0.78	0.86
750	0.69 ± 0.03	0.98 ± 0.04	0.80	0.88
800	0.79 ± 0.03	0.89 ± 0.03	0.81	0.89
850	0.82 ± 0.02	0.72 ± 0.03	0.83	0.90

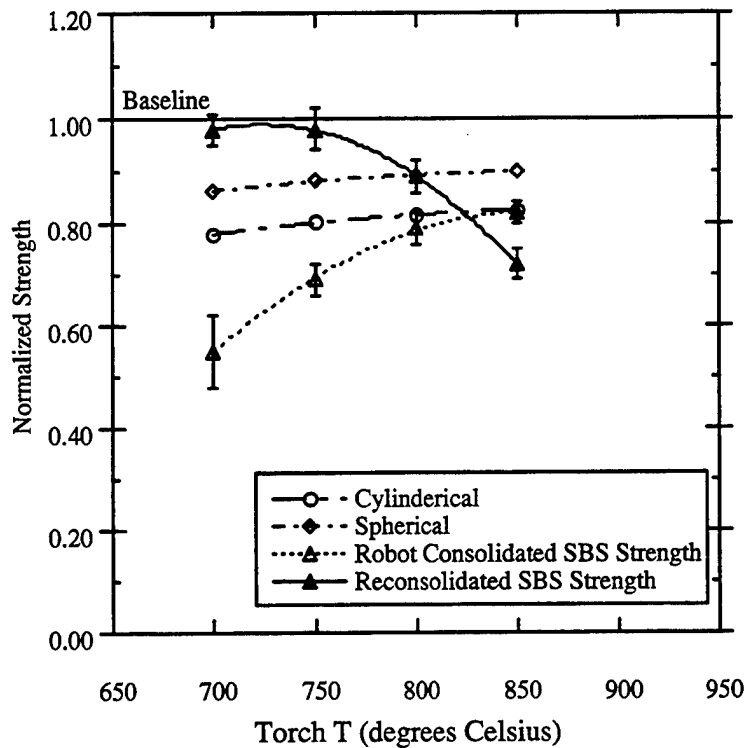


Figure 11. Effect of Void Content on SBS Strength for 20-mm/s Deposition Rate.

As can be seen from Figures 11 and 12, this analysis overestimates the ILSS, except for the torch temperatures of 800 and 850 °C for 20-mm/s deposition rate, indicating that the main mechanism contributing to the strength is interlaminar bonding. In fact, Bowles-type analyses assumes that the composite is a homogeneous material with voids and without any weak interface. For torch temperatures of 800 and 850 °C, polymer degradation occurring at the

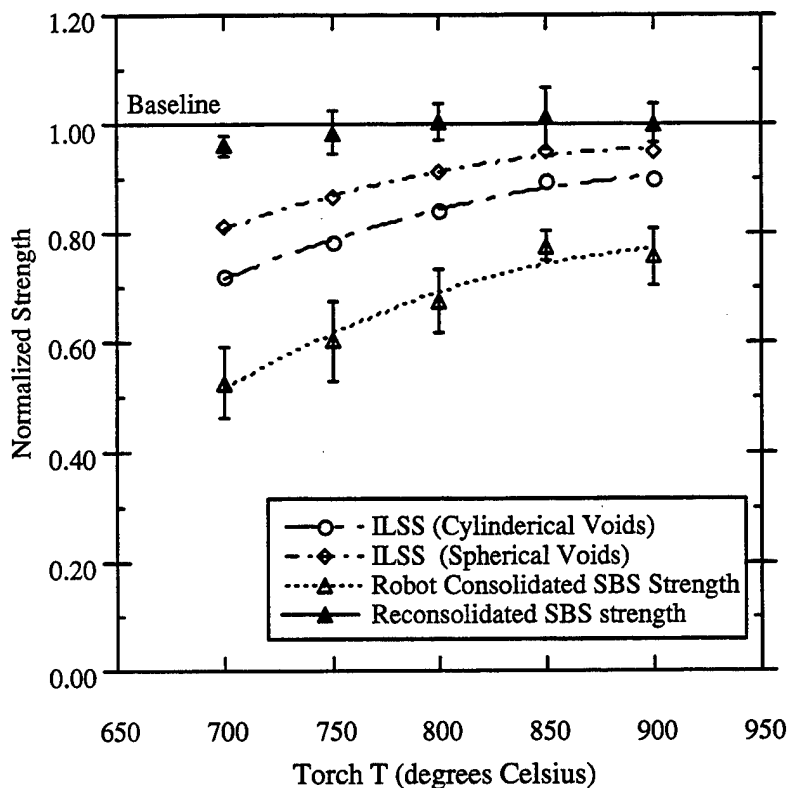


Figure 12. Effect of Void Content on SBS Strength for 40-mm/s Deposition Rate.

interface is believed to be the limiting factor for full-strength development, resulting in a strength loss that is not recoverable upon reconsolidation. Figures 13 and 14 show a fractured SBS specimen manufactured with settings of 700 °C torch temperatures and 20-mm/s deposition rate and 850 °C torch temperature and 20-mm/s deposition rate, respectively. For the former case, a single crack jumping from one interface to another is observed. For the latter case, the crack density was higher, but the cracks were still located at the interface and were separated from each other with a distance of one tow thickness.

4. Conclusions

Effect of processing variables on the quality of the coupons consolidated by thermoplastic fiber-placement method is investigated experimentally, in terms of strength and void content. As a general rule, at higher torch temperatures and slower deposition velocities, higher SBS

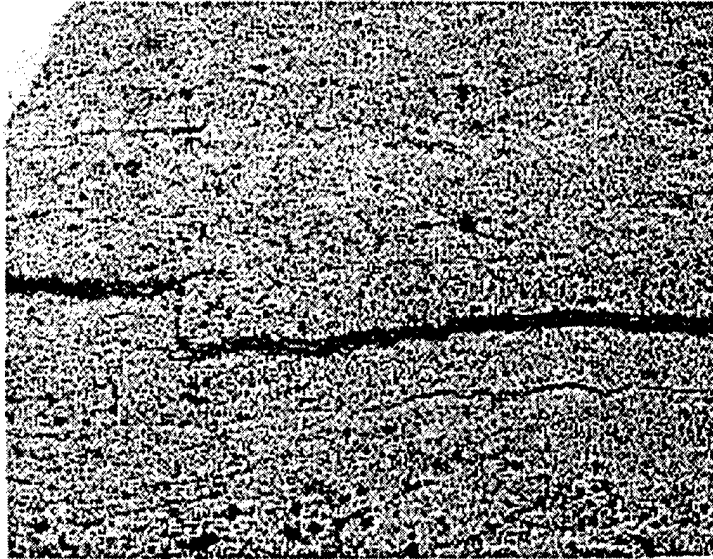


Figure 13. Micrograph of a Typical SBS Specimen Consolidated at 700 °C Torch Temperatures and 20-mm/s Deposition Rate.

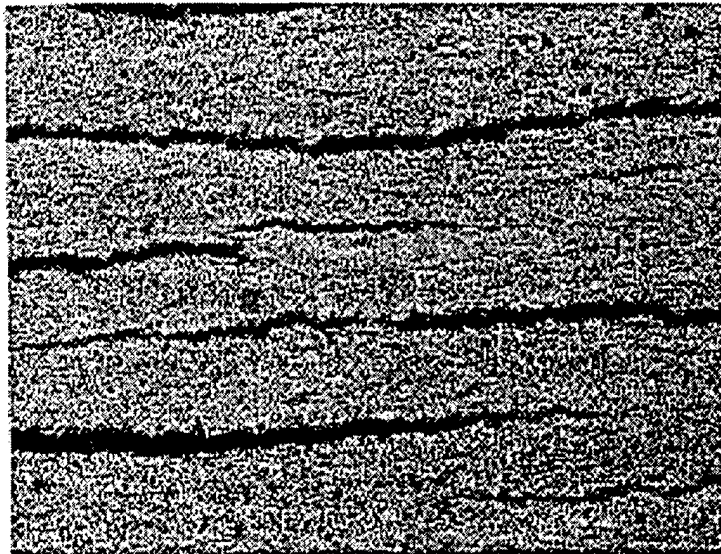


Figure 14. Micrograph of a Typical SBS Specimen Consolidated at 850 °C Torch Temperatures and 20-mm/s Deposition Rate.

strengths were obtained. Void content decreases with increasing torch temperature. However, at high torch temperatures and long dwell times, the “tradeoff” of polymer degradation prevents full-strength development during the process.

INTENTIONALLY LEFT BLANK.

5. References

1. Wells, G. M., and K. V. Steiner. "Robotic Thermoplastic Composite Placement." *Proceedings of 46th Annual Conference & Expo '91 of SPI Composites Institute*, Washington, DC, February 1991.
2. Felderhoff, K. D., and K. V. Steiner. "Development of a Compact Robotic Thermoplastic Fiber Placement Head." *Thirty-Sixth International SAMPE Symposium and Exhibition*, SAMPE, Anaheim, CA, 1993.
3. Fraude, E., and K. V. Steiner. "Investigation and Design of a Cut and Restart Mechanism for the Robotic Tape Placement Process." CCM Technical Report 93-38, University of Delaware Center for Composite Materials, 1993.
4. American Society for Testing of Materials. "Apparent Interlaminar Shear Strength for Parallel Fiber Composites by Short Beam Shear Method." ASTM D 2344-84, 1995.
5. Greszczuk, L. B. "Effect of Voids on Strength Properties of Filamentary Composites." *Proceedings, 22nd Annual Meeting of the Reinforced Plastics Division of the Society of Plastic Industry*, pp. 20-A.1-20-A.10, 1967.
6. Bowles, K. J., and S. Frimpong. "Void Effects on the Interlaminar Shear Strength of Unidirectional Graphite-Fiber-Reinforced Composites." *Journal of Composite Materials*, vol. 26, no. 10, pp. 1487-1509, 1992.
7. Olson, B. D., M. A. Lamontia, J. W. Gillespie, Jr., and T. A. Bogetti. "The Effects and Non-Destructive Evaluation of Defects in Thermoplastic Compression Loaded Composite Cylinders." *Journal of Thermoplastic Composites*, vol. 8, pp. 109-136, 1995.

INTENTIONALLY LEFT BLANK.

Appendix A:

**Thermal Gravimetric Analysis of Degradation of
Polyetherketoneketone (PEKK)**

INTENTIONALLY LEFT BLANK.

Thermal gravimetric analysis (TGA) was used to determine the temperature range in which the weight-loss mechanism is dominant. Figure A-1 shows the TGA diagrams of polyetherketoneketone (PEKK) films, as-received and processed at 900 °C torch temperature and 20-mm/s deposition rate. TGA tests were run both at air and nitrogen atmosphere. The weight-loss mechanism is not effective until a temperature of 550 °C, and only 0.4% of the weight is converted at a temperature of 450 °C. In order to delineate the effect of carbon fibers on degradation behavior, AS4/PEKK composite is also subjected to TGA tests, both in air and nitrogen atmosphere. TGA diagrams of AS4/PEKK composite are shown in Figure A-2. Comparing Figures A-1 and A-2, it can be asserted that the presence of carbon fibers does not affect the general behavior of degradation and the temperature at weight-loss starts.

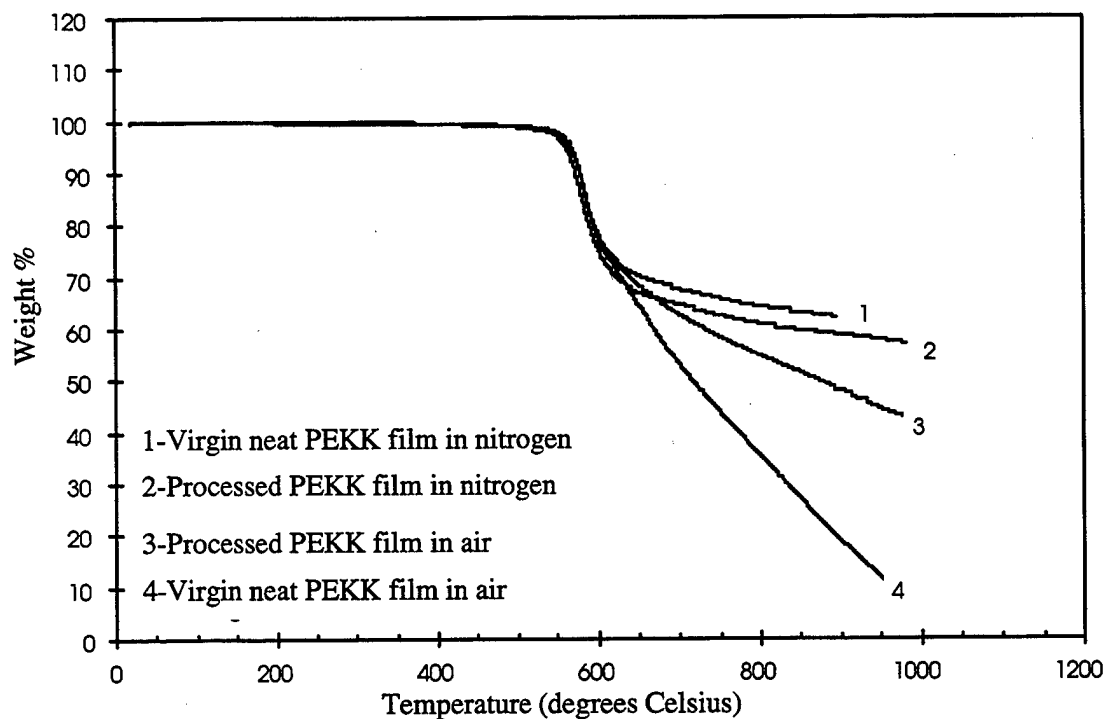


Figure A-1. TGA Diagram of PEKK Neat Films.

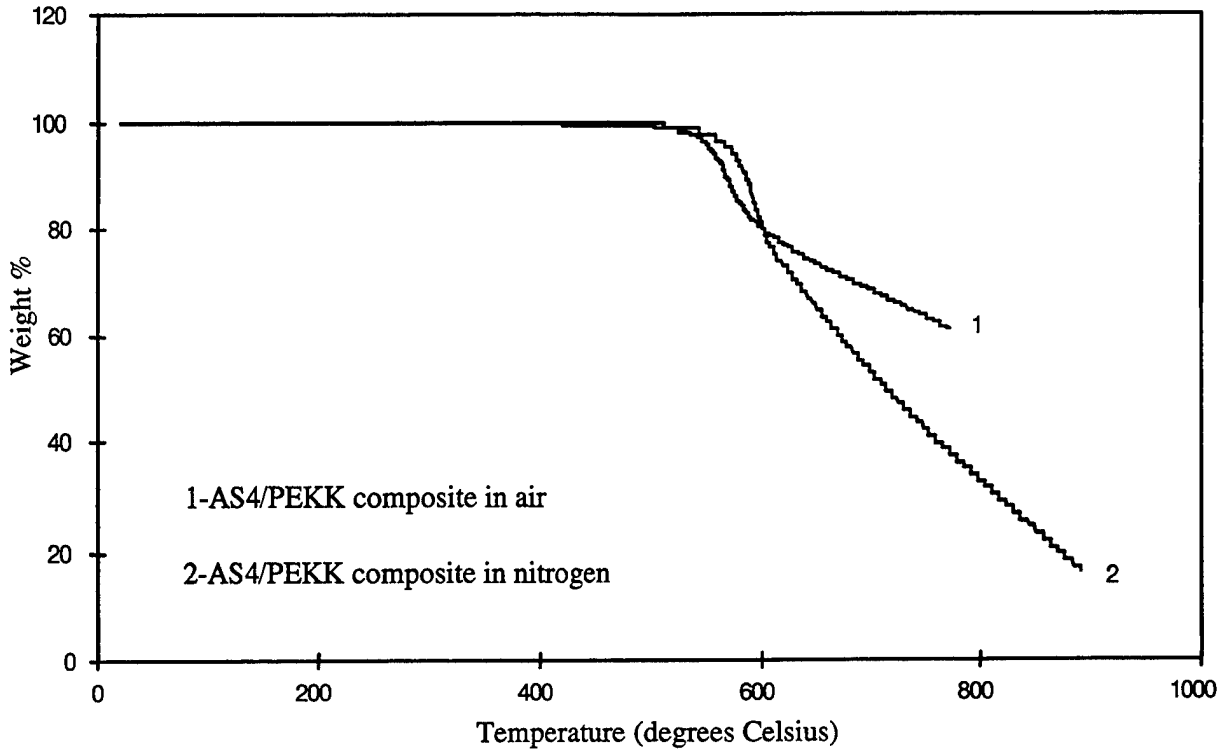


Figure A-2. TGA Diagram of AS4/PEKK Composite.

Figure A-3 shows the actual temperature profiles of the tow surface at 20-mm/s deposition rate for various torch temperature settings, recorded by an AGEMA Thermovision 900 thermal camera. Figure A-4 shows the model-predicted temperature profiles for various torch temperatures. Model-predicted and actual maximum temperatures are tabulated in Table A-1 and plotted vs. torch temperature settings in Figure A-5. Comparing model-predicted and actual maximum surface temperatures, it can be seen that our process simulation overestimates the maximum temperature. The maximum recorded temperature was 454 °C, which is far below the temperature at which weight loss is a dominant degradation mechanism.

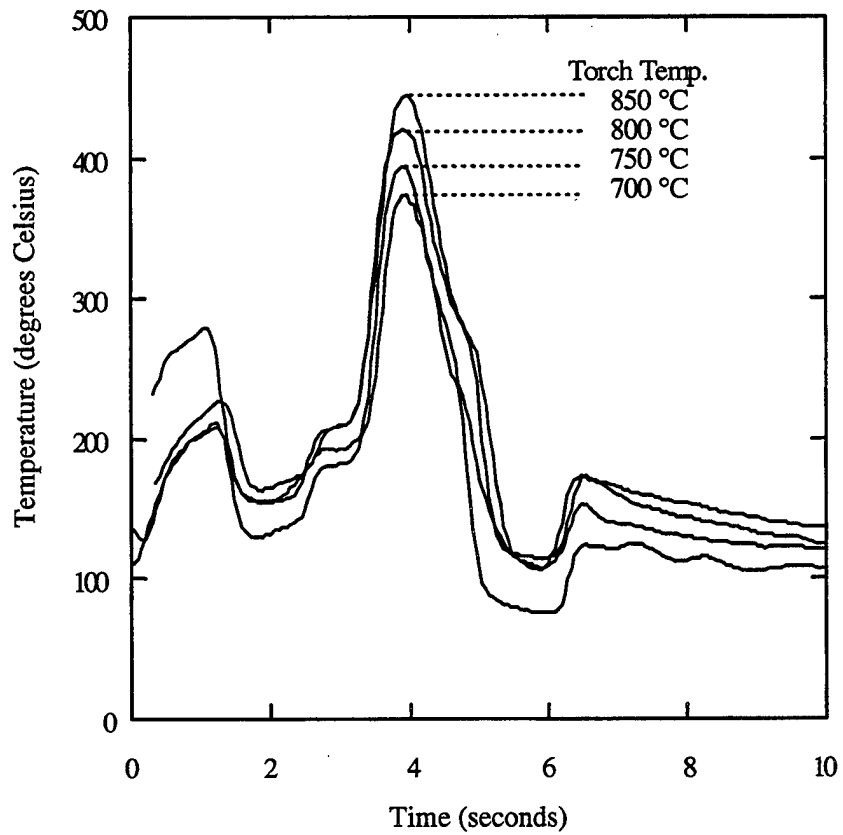


Figure A-3. Infrared Camera Readings of Surface Temperature at 20-mm/s Deposition Rate.

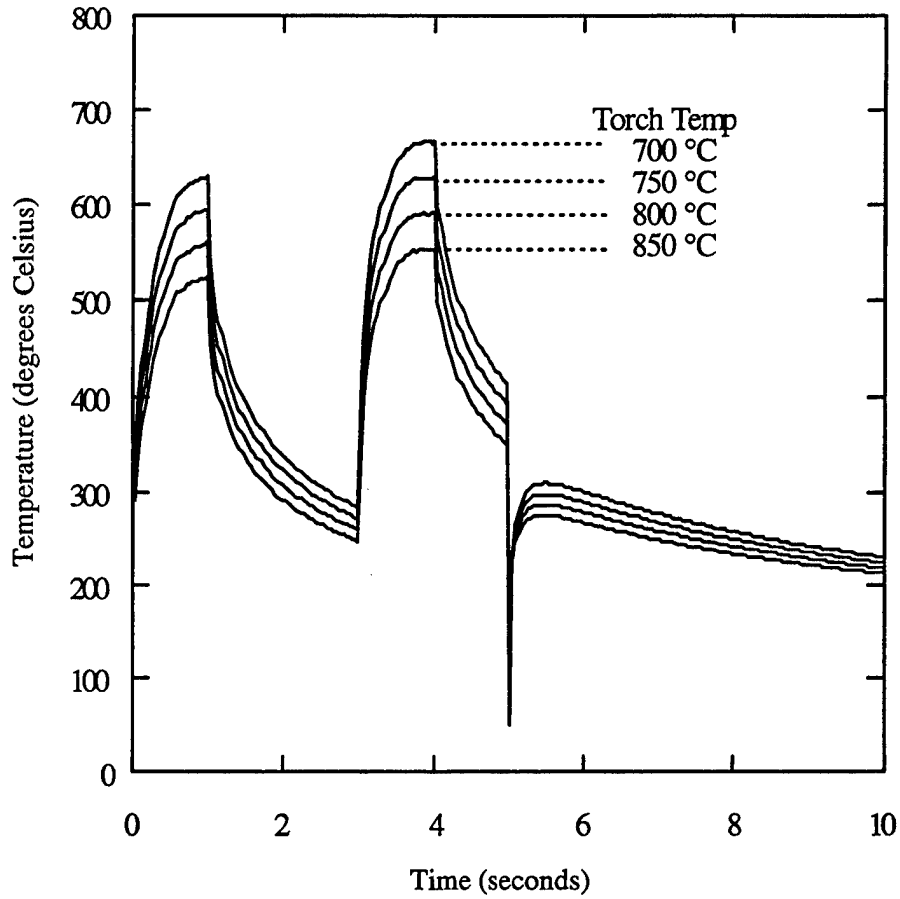


Figure A-4. Model Predictions for Surface Temperature at 20-mm/s Deposition Rate.

Table A-1. Maximum Surface Temperatures for Various Torch Temperature Settings at 20-mm/s Deposition Rate

Torch Temperature Setting (degrees Celsius)	Infrared Camera Reading (degrees Celsius)	Model Prediction (degrees Celsius)
700.00	373.3	553.4
750.00	394.2	591.5
800.00	419.8	629.6
850.00	445.4	667.7

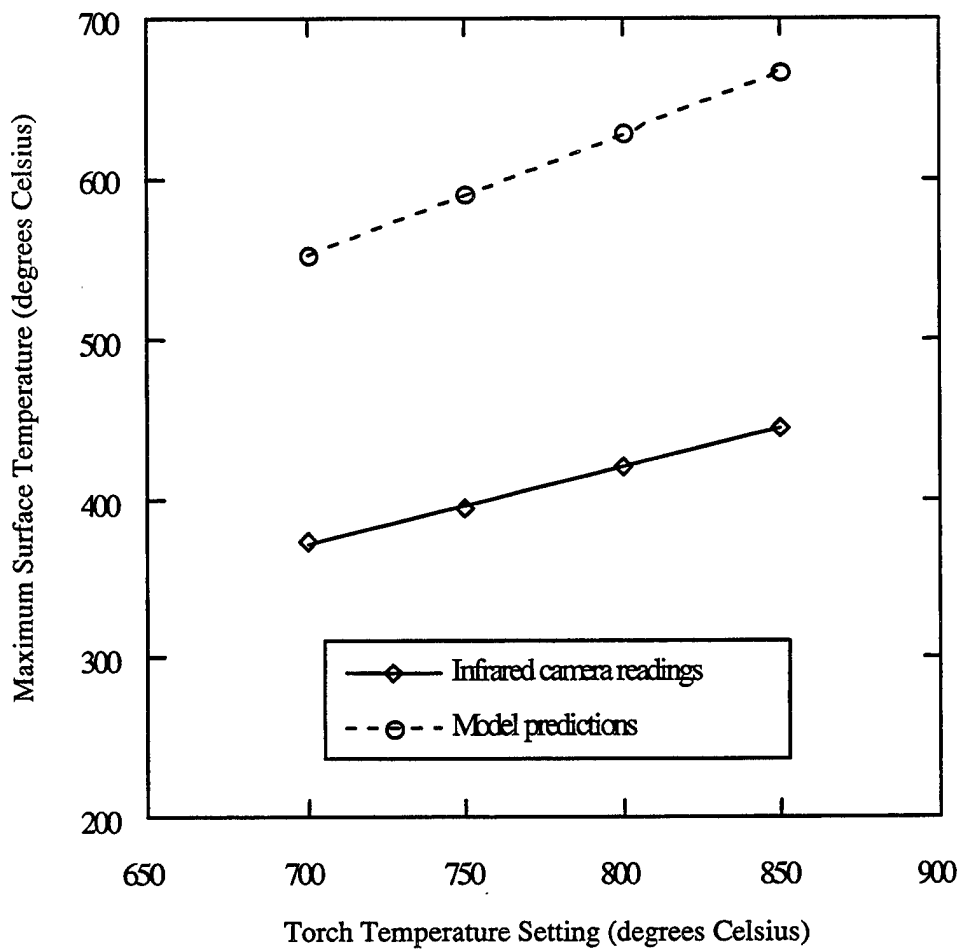


Figure A-5. Maximum Surface Temperature for Various Torch Temperature Settings at 20-mm/s Deposition Rate.

INTENTIONALLY LEFT BLANK.

Appendix B:

Fourier Transform Infrared Spectroscopic Study of Thermal Degradation of AS4/Polyetherketoneketone (PEKK) Composite

INTENTIONALLY LEFT BLANK.

Diffused reflectance Fourier-transform infrared spectroscopy (DRFTIR) has been used by Cole and Casella¹ to study the thermal degradation in polyetheretherketone (PEEK) carbon composites. Here, the same method is adopted to study the thermal degradation of AS4/polyetheretherketone (PEKK) composite during the tow placement process. Although diffused reflection spectra provide more limited information than a.t.r. spectra, they are, nevertheless, useful for following changes that occur upon thermal degradation at the surface of the composite.

Samples were prepared by processing tows of AS4/PEKK composite on a fiber-placement robot. However, for a single pass, no change in the diffuse reflectance spectra was observed and, therefore, the tow was forced to degrade by multiple passes of the fiber-placement head. Deposition velocity was held constant at 20 mm/s, and the torch temperature was varied to be 700, 800, and 900 °C. DRFTIR spectra were measured on a Nicolet 170SX instrument equipped with a "diffuse reflectance" fixture. The sample was mounted with its surface in a horizontal plane and oriented with the fiber direction parallel to the beam direction of the instrument. For each spectrum, 512 scans were accumulated at a resolution of 4 cm⁻¹. For qualitative treatment, the spectra were converted to Kubelka-Munk units. They were then baseline-corrected to remove the underlying carbon absorption. A baseline was drawn through the 1,750, 1,520, 1,460, 1,360, 1,040, 820, and 560 cm⁻¹ valleys. As the polymer degradation increased upon multiple passes, the overall intensity of the spectra decreased; hence, for qualitative comparisons, each spectra was integrated and scaled with respect to the reference spectrum of the undegraded tow.

Diffuse reflectance spectra for 700, 800, and 900 °C torch temperatures corrected for baseline and overall intensity are shown in Figures B-1, B-2, and B-3. The spectra are shifted vertically for better visualization. As can be seen from the figures, a new species in carbonyl

¹ Cole, K. C., and I. G. Casella. "Fourier Transform Infra-Red Spectroscopic Study of Thermal Degradation in Poly(Ether-Ether Ketone)-Carbon Composites." *Polymer*, vol. 34, no. 4, pp. 740-745, 1993.

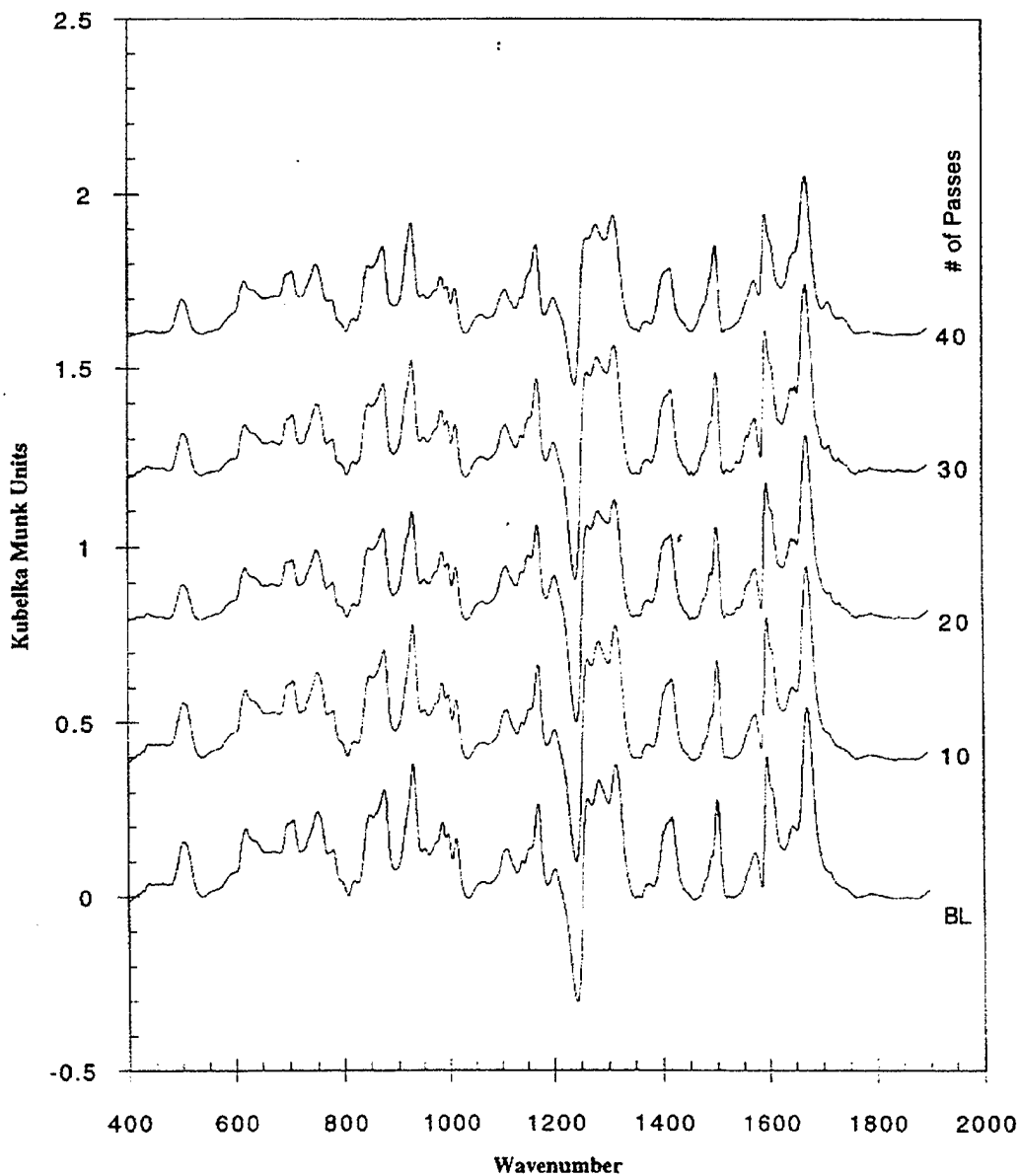


Figure B-1. Diffuse Reflectance Spectra for 700 °C Torch Temperature.

region ($1,716\text{ cm}^{-1}$) grows with increasing number of passes, and the rate of growth of this peak is higher for higher torch temperatures. Another change occurring in the spectra is the reduction in the height of 500 cm^{-1} peak. The origins of these species are uncertain. Cole and Casella¹

¹Cole, K. C., and I. G. Casella. "Fourier Transform Infra-Red Spectroscopic Study of Thermal Degradation in Poly(Ether-Ether Ketone)-Carbon Composites." *Polymer*, vol. 34, no. 4, pp. 740-745, 1993.

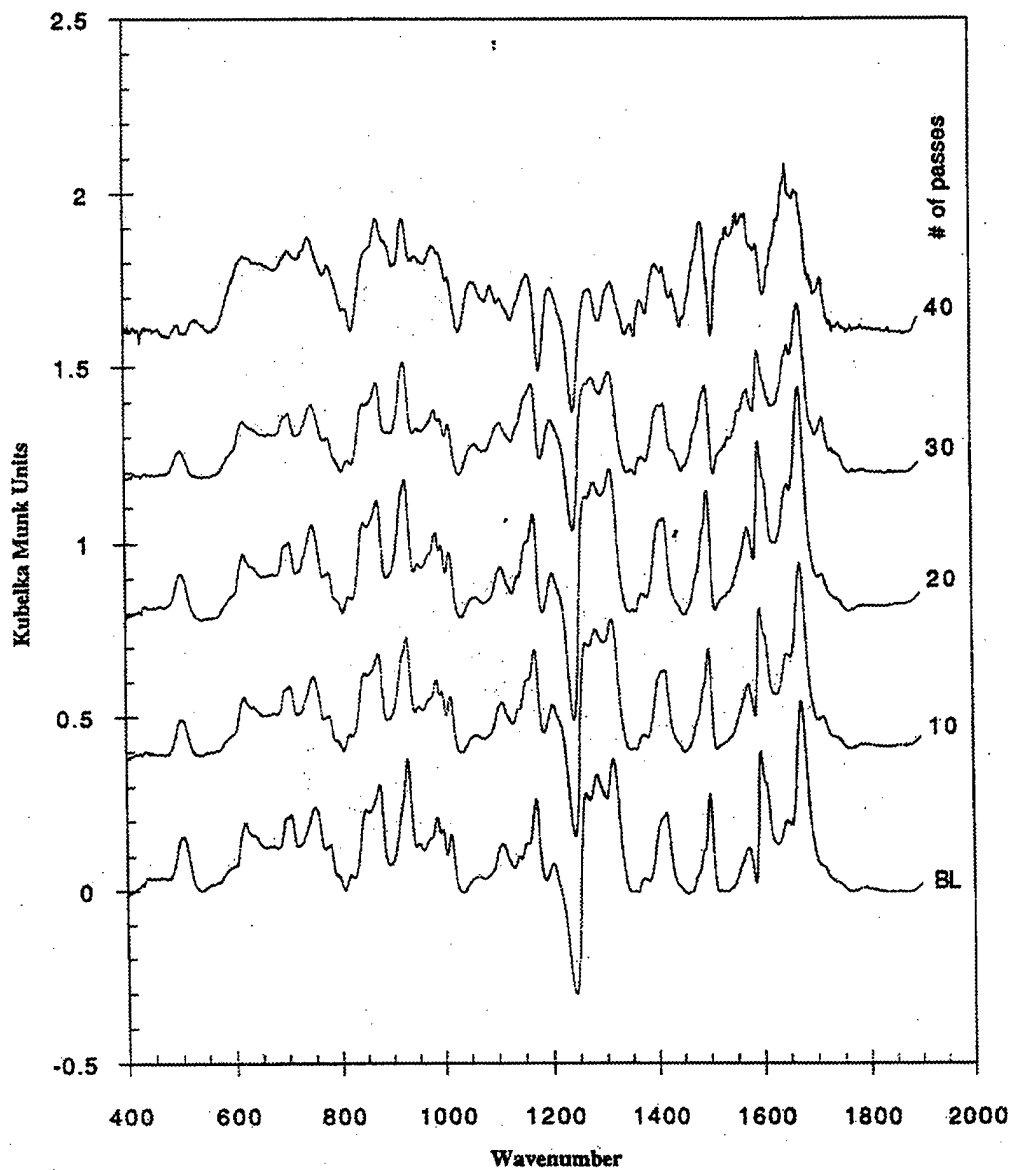


Figure B-2. Diffuse Reflectance Spectra for 800 °C Torch Temperature.

found a similar growing peak at $1,711\text{ cm}^{-1}$ for degradation of PEEK and proposed that this peak may result from cyclization of a diradical. Cyclization may reduce chain mobility and, hence, the ability to bond the polymer.

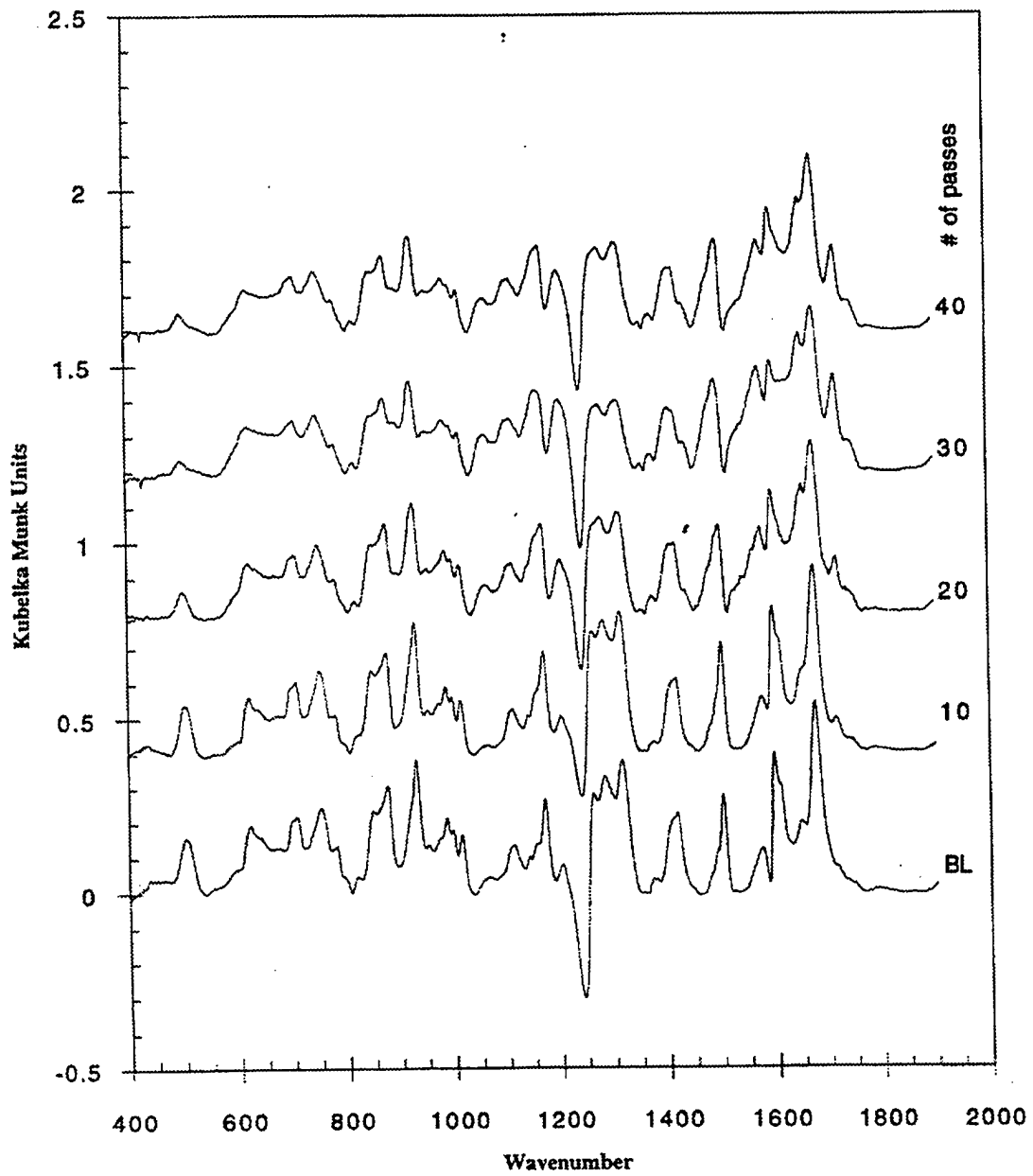


Figure B-3. Diffuse Reflectance Spectra for 900 °C Torch Temperature.

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	DEFENSE TECHNICAL INFORMATION CENTER DTIC DDA 8725 JOHN J KINGMAN RD STE 0944 FT BELVOIR VA 22060-6218
1	HQDA DAMO FDT 400 ARMY PENTAGON WASHINGTON DC 20310-0460
1	OSD OUSD(A&T)/ODDDR&E(R) R J TREW THE PENTAGON WASHINGTON DC 20301-7100
1	DPTY CG FOR RDA US ARMY MATERIEL CMD AMCRDA 5001 EISENHOWER AVE ALEXANDRIA VA 22333-0001
1	INST FOR ADVNCD TCHNLGY THE UNIV OF TEXAS AT AUSTIN PO BOX 202797 AUSTIN TX 78720-2797
1	DARPA B KASPAR 3701 N FAIRFAX DR ARLINGTON VA 22203-1714
1	NAVAL SURFACE WARFARE CTR CODE B07 J PENNELLA 17320 DAHLGREN RD BLDG 1470 RM 1101 DAHLGREN VA 22448-5100
1	US MILITARY ACADEMY MATH SCI CTR OF EXCELLENCE DEPT OF MATHEMATICAL SCI MADN MATH THAYER HALL WEST POINT NY 10996-1786

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	DIRECTOR US ARMY RESEARCH LAB AMSRL D D R SMITH 2800 POWDER MILL RD ADELPHI MD 20783-1197
1	DIRECTOR US ARMY RESEARCH LAB AMSRL DD 2800 POWDER MILL RD ADELPHI MD 20783-1197
1	DIRECTOR US ARMY RESEARCH LAB AMSRL CS AS (RECORDS MGMT) 2800 POWDER MILL RD ADELPHI MD 20783-1145
3	DIRECTOR US ARMY RESEARCH LAB AMSRL CI LL 2800 POWDER MILL RD ADELPHI MD 20783-1145
	<u>ABERDEEN PROVING GROUND</u>
4	DIR USARL AMSRL CI LP (BLDG 305)

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	DIRECTOR US ARMY RESEARCH LAB AMSRL CP CA D SNIDER 2800 POWDER MILL RD ADELPHI MD 20783-1145
1	DIRECTOR US ARMY RESEARCH LAB AMSRL OP SD TA 2800 POWDER MILL ROAD ADELPHI MD 20783-1145
3	DIRECTOR US ARMY RESEARCH LAB AMSRL OP SD TL 2800 POWDER MILL ROAD ADELPHI MD 20783-1145
1	DIRECTOR US ARMY RESEARCH LAB AMSRL OP SD TP 2800 POWDER MILL ROAD ADELPHI MD 20783-1145
2	DIRECTOR US ARMY RESEARCH LAB AMSRL OP CI AD TECH PUB BR RECORDS MGMT ADMIN 2800 POWDER MILL ROAD ADELPHI MD 20783-1197
1	HQDA DAMI FIT NOLAN BLDG WASHINGTON DC 20310-1025
1	DIRECTOR DA OASARDA SARD SO 103 ARMY PENTAGON WASHINGTON DC 20310-0103
1	DEPUTY ASST SCY FOR R&T SARD TT RM 3EA79 THE PENTAGON WASHINGTON DC 20301-7100

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	COMMANDER US ARMY MATERIEL CMD AMXMI INT 5001 EISENHOWER AVE ALEXANDRIA VA 22333-0001
2	COMMANDER US ARMY ARDEC AMSTA AR AE WW E BAKER J PEARSON PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY ARDEC AMSTA AR TD C SPINELLI PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY ARDEC AMSTA AR FSE T GORA PICATINNY ARSENAL NJ
6	COMMANDER US ARMY ARDEC AMSTA AR CCH A W ANDREWS S MUSALLI R CARR M LUCIANO E LOGSDEN T LOUZEIRO PICATINNY ARSENAL NJ 07806-5000
4	COMMANDER US ARMY ARDEC AMSTA AR CC G PAYNE J GEHBAUER C BAULIEU H OPAT PICATINNY ARSENAL NJ 07806-5000

NO. OF
COPIES ORGANIZATION

1 COMMANDER
US ARMY ARDEC
AMSTA AR CCH P
J LUTZ
PICATINNY ARSENAL NJ
07806-5000

1 COMMANDER
US ARMY ARDEC
AMSTA AR FSF T
C LIVECCHIA
PICATINNY ARSENAL NJ
07806-5000

1 COMMANDER
US ARMY ARDEC
AMSTA AR QAC T C
C PATEL
PICATINNY ARSENAL NJ
07806-5000

2 COMMANDER
US ARMY ARDEC
AMSTA AR M
D DEMELLA
F DIORIO
PICATINNY ARSENAL NJ
07806-5000

3 COMMANDER
US ARMY ARDEC
AMSTA AR FSA
A WARNASH
B MACHAK
M CHIEFA
PICATINNY ARSENAL NJ
07806-5000

2 COMMANDER
US ARMY ARDEC
AMSTA AR FSP G
M SCHIKSNIS
D CARLUCCI
PICATINNY ARSENAL NJ
07806-5000

NO. OF
COPIES ORGANIZATION

1 COMMANDER
US ARMY ARDEC
AMSTA AR FSP A
P KISATSKY
PICATINNY ARSENAL NJ
07806-5000

2 COMMANDER
US ARMY ARDEC
AMSTA AR CCH C
H CHANIN
S CHICO
PICATINNY ARSENAL NJ
07806-5000

9 COMMANDER
US ARMY ARDEC
AMSTA AR CCH B
P DONADIA
F DONLON
P VALENTI
C KNUTSON
G EUSTICE
S PATEL
G WAGNECZ
R SAYER
F CHANG
PICATINNY ARSENAL NJ
07806-5000

6 COMMANDER
US ARMY ARDEC
AMSTA AR CCL
F PUZYCKI
R MCHUGH
D CONWAY
E JAROSZEWSKI
R SCHLENNER
M CLUNE
PICATINNY ARSENAL NJ
07806-5000

1 COMMANDER
US ARMY ARDEC
AMSTA AR QAC T
D RIGOGLIOSO
PICATINNY ARSENAL NJ
07806-5000

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
1	COMMANDER US ARMY ARDEC AMSTA AR SRE D YEE PICATINNY ARSENAL NJ 07806-5000	2	PEO FIELD ARTILLERY SYSTEMS SFAE FAS PM H GOLDMAN T MCWILLIAMS PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY ARDEC AMSTA AR WET T SACHAR BLDG 172 PICATINNY ARSENAL NJ 07806-5000	6	PM SADARM SFAE GCSS SD COL B ELLIS M DEVINE R KOWALSKI W DEMASSI J PRITCHARD S HROWNAK PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY ARDEC SMCAR ASF PICATINNY ARSENAL NJ 07806-5000	1	COMMANDER US ARMY ARDEC PRODUCTION BASE MODERN ACTY AMSMC PBM K PICATINNY ARSENAL NJ 07806-5000
1	COMMANDER US ARMY ARDEC AMSTA AR WEL F INTELLIGENCE SPECIALIST M GUERRIERE PICATINNY ARSENAL NJ 07806-5000	3	COMMANDER U S ARMY TACOM PM TACTICAL VEHICLES SFAE TVL SFAE TVM SFAE TVH 6501 ELEVEN MILE RD WARREN MI 48397-5000
11	PROJECT MANAGER TANK MAIN ARMAMENT SYSTEMS SFAE GSSC TMA R MORRIS C KIMKER D GUZOWICZ E KOPACZ R ROESER R DARCY R MCDANOLDS L D ULISSE C ROLLER J MCGREEN B PATTTER PICATINNY ARSENAL NJ 07806-5000	1	COMMANDER U S ARMY TACOM PM ABRAMS SFAE ASM AB 6501 ELEVEN MILE RD WARREN MI 48397-5000
		1	COMMANDER U S ARMY TACOM PM BFVS SFAE ASM BV 6501 ELEVEN MILE RD WARREN MI 48397-5000

NO. OF
COPIES ORGANIZATION

NO. OF
COPIES ORGANIZATION

1 COMMANDER
U S ARMY TACOM
PM AFAS
SFAE ASM AF
6501 ELEVEN MILE RD
WARREN MI 48397-5000

1 COMMANDER
U S ARMY TACOM
PM GROUND SYSTEMS
INTEGRATION
SFAE GCSS W GSI
R LABATILLE
6501 ELEVEN MILE RD
WARREN MI 48397-5000

2 COMMANDER
U S ARMY TACOM
PM SURV SYS
SFAE ASM SS
T DEAN
SFAE GCSS W GSI M
D COCHRAN
6501 ELEVEN MILE RD
WARREN MI 48397-5000

1 COMMANDER
U S ARMY TACOM
CHIEF ABRAMS TESTING
SFAE GCSS W AB QT
T KRASKIEWICZ
6501 ELEVEN MILE RD
WARREN MI 48397-5000

1 COMMANDER
U S ARMY TACOM
PM RDT&E
SFAE GCSS W AB
J GODELL
6501 ELEVEN MILE RD
WARREN MI 48397-5000

1 COMMANDER
US ARMY TACOM
AMSTA SF
WARREN MI 48397-5000

1 COMMANDER
U S ARMY TACOM
PM SURVIVABLE SYSTEMS
SFAE GCSS W GSI H
M RYZYI
6501 ELEVEN MILE RD
WARREN MI 48397-5000

1 COMMANDER
SMCWV QAE Q
B VANINA
BLDG 44
WATERVLIET ARSENAL
WATERVLIET NY 12189-4050

1 COMMANDER
U S ARMY TACOM
PM BFV
SFAE GCSS W BV
S DAVIS
6501 ELEVEN MILE RD
WARREN MI 48397-5000

14 COMMANDER
US ARMY TACOM
ASMTA TR R
J CHAPIN
R MCCLELLAND
D THOMAS
J BENNETT
D HANSEN
AMSTA JSK
S GOODMAN
J FLORENCE
K IYER
J THOMSON
AMSTA TR D
D OSTBERG
L HINOJOSA
B RAJU
AMSTA CS SF
H HUTCHINSON
F SCHWARZ
WARREN MI 48397-5000

1 COMMANDER
U S ARMY TACOM
PM LIGHT TACTICAL
VEHICLES
AMSTA TR S
AJ J MILLS MS 209
6501 ELEVEN MILE RD
WARREN MI 48397-5000

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	COMMANDER SMCWV SPM T MCCLOSKEY BLDG 253 WATERVLIET ARSENAL WATERVLIET NY 12189-4050
10	BENET LABORATORIES AMSTA AR CCB R FISCELLA G D ANDREA M SCAVULO G SPENCER P WHEELER K MINER J VASILAKIS G FRIAR R HASENBEIN SMCAR CCB R S SOPOK WATERVLIET NY 12189
2	TSM ABRAMS ATZK TS S JABURG W MEINSHAUSEN FT KNOX KY 40121
3	ARMOR SCHOOL ATTN ATZK TD R BAUEN J BERG A POMEY FT KNOX KY 40121
2	HQ IOC TANK AMMO TEAM AMSIO SMT R CRAWFORD W HARRIS ROCK ISLAND IL 61299-6000
1	DIRECTOR U S ARMY AMCOM SFAE AV RAM TV D CALDWELL BUILDING 5300 REDSTONE ARSENAL AL 35898

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
4	DIRECTOR US ARMY CECOM NIGHT VISION AND ELECTRONIC SENSORS DIRECTORATE AMSEL RD NV CM CCD R ADAMS R MCLEAN A YINGST AMSEL RD NV VISP E JACOBS 10221 BURBECK RD FT BELVOIR VA 22060-5806
2	CDR USA AMCOM AVIATION APPLIED TECH DIR J SCHUCK FT EUSTIS VA 23604-5577
1	U S ARMY CRREL P DUTTA 72 LYME RD HANOVER NH 03755
1	US ARMY CERL R LAMPO 2902 NEWMARK DR CHAMPAIGN IL 61822
2	U S ARMY CORP OF ENGINEERS CERD C T LIU CEW ET T TAN 20 MASS AVE NW WASHINGTON DC 20314
10	DIRECTOR US ARMY NATL GRND INTEL CTR D LEITER S EITELMAN M HOLTUS M WOLFE S MINGLEDORF H C ARDLEIGH J GASTON W GSTATTENBAUER R WARNER J CRIDER 220 SEVENTH STREET NE CHARLOTTESVILLE VA 22091

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
6	US ARMY SBCCOM SOLDIER SYSTEMS CENTER BALLISTICS TEAM J WARD MARINE CORPS TEAM J MACKIEWICZ BUS AREA ADVOCACY TEAM W HASKELL SSCNC WST W NYKVIST T MERRILL S BEAUDOIN KANSAS ST NATICK MA 01760-5019
1	US ARMY COLD REGIONS RSCH & ENGRNG LAB P DUTTA 72 LYME RD HANOVER NH 03755
1	SYSTEM MANAGER ABRAMS BLDG 1002 RM 110 ATZK TS LTC J H NUNN FT KNOX KY 40121
9	US ARMY RESEARCH OFFICE A CROWSON J CHANDRA H EVERETT J PRATER R SINGLETON G ANDERSON D STEPP D KISEROW J CHANG PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211
1	DIRECTORATE OF CMBT DEVELOPMENT C KJORO 320 ENGINEER LOOP STE 141 FT LEONARD WOOD MO 65473-8929

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	COMMANDANT U S ARMY FIELD ARTILLERY CTR AT FT SILL ATFS CD LTC BUMGARNER FT SILL OK 73503 5600
1	CHIEF USAIC LTC T J CUMMINGS ATZB COM FT BENNING GA 31905-5800
1	NAVAL AIR SYSTEMS CMD J THOMPSON 48142 SHAW RD UNIT 5 PATUXENT RIVER MD 20670
1	NAVAL SURFACE WARFARE CTR DAHLGREN DIV CODE G06 DAHLGREN VA 22448
1	NAVAL SURFACE WARFARE CTR TECH LIBRARY CODE 323 17320 DAHLGREN RD DAHLGREN VA 22448
3	NAVAL RESEARCH LAB I WOLOCK CODE 6383 R BADALIANCE CODE 6304 L GAUSE WASHINGTON DC 20375
1	NAVAL SURFACE WARFARE CTR CRANE DIVISION M JOHNSON CODE 20H4 LOUISVILLE KY 40214-5245
2	COMMANDER NAVAL SURFACE WARFARE CTR CADEROCK DIVISION R PETERSON CODE 2020 M CRITCHFIELD CODE 1730 BETHESDA MD 20084
2	NAVAL SURFACE WARFARE CTR U SORATHIA C WILLIAMS CD 6551 9500 MACARTHUR BLVD WEST BETHESDA MD 20817

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	DAVID TAYLOR RESEARCH CTR SHIP STRUCTURES & PROTECTION DEPARTMENT CODE 1702 J CORRADO BETHESDA MD 20084	1	NAVSEA OJRI PEO DD21 PMS500 G CAMPONESCHI 2351 JEFFERSON DAVIS HWY ARLINGTON VA 22242-5165
2	DAVID TAYLOR RESEARCH CTR R ROCKWELL W PHYLLAIER BETHESDA MD 20054-5000	1	EXPEDITIONARY WARFARE DIV N85 F SHOUP 2000 NAVY PENTAGON WASHINGTON DC 20350-2000
1	OFFICE OF NAVAL RESEARCH D SIEGEL CODE 351 800 N QUINCY ST ARLINGTON VA 22217-5660	1	AFRL MLBC 2941 P STREET RM 136 WRIGHT PATTERSON AFB OH 45433-7750
8	NAVAL SURFACE WARFARE CTR J FRANCIS CODE G30 D WILSON CODE G32 R D COOPER CODE G32 J FRAYSSE CODE G33 E ROWE CODE G33 T DURAN CODE G33 L DE SIMONE CODE G33 R HUBBARD CODE G33 DAHLGREN VA 22448	1	AFRL MLSS R THOMSON 2179 12TH STREET RM 122 WRIGHT PATTERSON AFB OH 45433-7718
1	NAVAL SEA SYSTEMS CMD D LIESE 2531 JEFFERSON DAVIS HIGHWAY ARLINGTON VA 22242-5160	2	AFRL F ABRAMS J BROWN BLDG 653 2977 P STREET STE 6 WRIGHT PATTERSON AFB OH 45433-7739
1	NAVAL SURFACE WARFARE M LACY CODE B02 17320 DAHLGREN RD DAHLGREN VA 22448	1	AFRL MLS OL 7278 4TH STREET BLDG 100 BAY D L COULTER HILL AFB UT 84056-5205
1	OFFICE OF NAVAL RES J KELLY 800 NORTH QUINCEY ST ARLINGTON VA 22217-5000	1	OSD JOINT CCD TEST FORCE OSD JCCD R WILLIAMS 3909 HALLS FERRY RD VICKSBURG MS 29180-6199
2	NAVAL SURFACE WARFARE CTR CARDEROCK DIVISION R CRANE CODE 2802 C WILLIAMS CODE 6553 3A LEGGETT CIR BETHESDA MD 20054-5000	1	DEFENSE NUCLEAR AGENCY INNOVATIVE CONCEPTS DIV R ROHR 6801 TELEGRAPH RD ALEXANDRIA VA 22310-3398

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	WATERWAYS EXPERIMENT D SCOTT 3909 HALLS FERRY RD SC C VICKSBURG MS 39180
3	DARPA M VANFOSSSEN S WAX L CHRISTODOULOU 3701 N FAIRFAX DR ARLINGTON VA 22203-1714
2	SERDP PROGRAM OFC PM P2 C PELLERIN B SMITH 901 N STUART ST SUITE 303 ARLINGTON VA 22203
1	FAA MIL HDBK 17 CHAIR L ILCEWICZ 1601 LIND AVE SW ANM 115N RENTON VA 98055
2	FAA TECH CENTER D OPLINGER AAR 431 P SHYPRYKEVICH AAR 431 ATLANTIC CITY NJ 08405
1	OFC OF ENVIRONMENTAL MGMT U S DEPT OF ENERGY P RITZCOVAN 19901 GERMANTOWN RD GERMANTOWN MD 20874-1928
1	LOS ALAMOS NATL LAB F ADDESSIO MS B216 PO BOX 1633 LOS ALAMOS NM 87545
1	OAK RIDGE NATL LAB R M DAVIS PO BOX 2008 OAK RIDGE TN 37831-6195

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
5	DIRECTOR LAWRENCE LIVERMORE NATL LAB R CHRISTENSEN S DETERESA F MAGNESS M FINGER MS 313 M MURPHY L 282 PO BOX 808 LIVERMORE CA 94550
7	NIST R PARNAS J DUNKERS M VANLANDINGHAM MS 8621 J CHIN MS 8621 D HUNSTON MS 8543 J MARTIN MS 8621 D DUTHINH MS 8611 100 BUREAU DR GAITHERSBURG MD 20899
1	OAK RIDGE NATL LAB C EBERLE MS 8048 PO BOX 2009 OAK RIDGE TN 37831
1	OAK RIDGE NATL LAB C D WARREN MS 8039 PO BOX 2009 OAK RIDGE TN 37922
4	DIRECTOR SANDIA NATL LABS APPLIED MECHANICS DEPT DIVISION 8241 W KAWAHARA K PERANO D DAWSON P NIELAN PO BOX 969 LIVERMORE CA 94550-0096
1	LAWRENCE LIVERMORE NATIONAL LAB M MURPHY PO BOX 808 L 282 LIVERMORE CA 94550

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
3	NASA LANGLEY RESEARCH CTR MS 266 AMSRL VS W ELBER F BARTLETT JR G FARLEY HAMPTON VA 23681-0001
1	NASA LANGLEY RESEARCH CTR T GATES MS 188E HAMPTON VA 23661-3400
1	USDOT FEDERAL RAILROAD RDV 31 M FATEH WASHINGTON DC 20590
1	DOT FHWA J SCALZI 400 SEVENTH ST SW 3203 HNG 32 WASHINGTON DC 20590
1	FHWA E MUNLEY 6300 GEORGETOWN PIKE MCLEAN VA 22101
1	CENTRAL INTELLIGENCE AGENCY OTI WDAG GT W L WALTMAN PO BOX 1925 WASHINGTON DC 20505
1	MARINE CORPS INTEL ACTY D KOSITZKE 3300 RUSSELL RD SUITE 250 QUANTICO VA 22134-5011
1	NATL GRND INTELLIGENCE CTR DIRECTOR IANG TMT 220 SEVENTH ST NE CHARLOTTESVILLE VA 22902-5396
1	DIRECTOR DEFENSE INTELLIGENCE AGENCY TA 5 K CRELLING WASHINGTON DC 20310

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	GRAPHITE MASTERS INC J WILLIS 3815 MEDFORD ST LOS ANGELES CA 90063-1900
1	ADVANCED GLASS FIBER YARNS T COLLINS 281 SPRING RUN LN STE A DOWNINGTON PA 19335
1	COMPOSITE MATERIALS INC D SHORTT 19105 63 AVE NE PO BOX 25 ARLINGTON WA 98223
1	COMPOSITE MATERIALS INC R HOLLAND 11 JEWEL COURT ORINDA CA 94563
1	COMPOSITE MATERIALS INC C RILEY 14530 S ANSON AVE SANTA FE SPRINGS CA 90670
2	COMPOSIX D BLAKE L DIXON 120 O NEILL DR HEBRUN OHIO 43025
4	CYTEC FIBERITE R DUNNE D KOHLI M GILLIO R MAYHEW 1300 REVOLUTION ST HAVRE DE GRACE MD 21078
2	SIMULA J COLTMAN R HUYETT 10016 S 51ST ST PHOENIX AZ 85044
1	SIOUX MFG B KRIEL PO BOX 400 FT TOTTEN ND 58335

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	PROTECTION MATERIALS INC M MILLER F CRILLEY 14000 NW 58 CT MIAMI LAKES FL 33014
3	FOSTER MILLER J J GASSNER M ROYLANCE W ZUKAS 195 BEAR HILL RD WALTHAM MA 02354-1196
1	ROM DEVELOPMENT CORP R O MEARA 136 SWINEBURNE ROW BRICK MARKET PLACE NEWPORT RI 02840
2	TEXTRON SYSTEMS T FOLTZ M TREASURE 201 LOWELL ST WILMINGTON MA 08870-2941
1	JPS GLASS L CARTER PO BOX 260 SLATER RD SLATER SC 29683
1	O GARA HESS & EISENHARDT M GILLESPIE 9113 LESAINTE DR FAIRFIELD OH 45014
2	MILLIKEN RESEARCH CORP H KUHN M MACLEOD PO BOX 1926 SPARTANBURG SC 29303
1	CONNEAUGHT INDUSTRIES INC J SANTOS PO BOX 1425 COVENTRY RI 02816

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	BATTELLE C R HARGREAVES 505 KING AVE COLUMBUS OH 43201-2681
2	BATTELLE NATICK OPERATIONS J CONNORS B HALPIN 209 W CENTRAL ST STE 302 NATICK MA 01760
1	BATTELLE NW DOE PNNL T HALL MS K231 BATTELLE BLVD RICHLAND WA 99352
3	PACIFIC NORTHWEST LAB M SMITH G VAN ARSDALE R SHIPPELL PO BOX 999 RICHLAND WA 99352
1	ARMTEC DEFENSE PRODUCTS S DYER 85 901 AVE 53 PO BOX 848 COACHELLA CA 92236
2	ADVANCED COMPOSITE MATLS CORP P HOOD J RHODES 1525 S BUNCOMBE RD GREER SC 29651-9208
2	GLCC INC J RAY M BRADLEY 103 TRADE ZONE DR STE 26C WEST COLUMBIA SC 29170
2	AMOCO PERFORMANCE PRODUCTS M MICHNO JR J BANISAUKAS 4500 MCGINNIS FERRY RD ALPHARETTA GA 30202-3944

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
1	SAIC M PALMER 2109 AIR PARK RD S E ALBUQUERQUE NM 87106	1	PROJECTILE TECHNOLOGY INC 515 GILES ST HAVRE DE GRACE MD 21078
1	SAIC ATTN G CHRYSSOMALLIS 3800 W 80TH ST STE 1090 BLOOMINGTON MN 55431	1	CUSTOM ANALYTICAL ENG SYS INC A ALEXANDER 1300 TENSOR LN NE FLINTSTONE MD 21530
1	AAI CORPORATION DR T G STASTNY PO BOX 126 HUNT VALLEY MD 21030-0126	2	LORAL VOUGHT SYSTEMS G JACKSON K COOK 1701 W MARSHALL DR GRAND PRAIRIE TX 75051
1	JOHN HEBERT PO BOX 1072 HUNT VALLEY MD 21030-0126	5	AEROJET GEN CORP D PILLASCH T COULTER C FLYNN D RUBAREZUL M GREINER 1100 WEST HOLLYVALE ST AZUSA CA 91702-0296
12	ALLIANT TECHSYSTEMS INC C CANDLAND C AAKHUS R BECKER B SEE N VLAHAKUS R DOHRN S HAGLUND D FISHER W WORRELL R COPENHAFER M HISSONG D KAMDAR 600 2ND ST NE HOPKINS MN 55343-8367	3	HEXCEL INC R BOE F POLICELLI J POESCH PO BOX 98 MAGNA UT 84044
3	ALLIANT TECHSYSTEMS INC J CONDON E LYNAM J GERHARD WV01 16 STATE RT 956 PO BOX 210 ROCKET CENTER WV 26726-0210	3	HERCULES INC G KUEBELER J VERMEYCHUK B MANDERVILLE JR HERCULES PLAZA WILMINGTON DE 19894
1	APPLIED COMPOSITES W GRISCH 333 NORTH SIXTH ST ST CHARLES IL 60174	1	BRIGS COMPANY J BACKOFEN 2668 PETERBOROUGH ST HERNDON VA 22071-2443
		1	ZERNOW TECHNICAL SERVICES L ZERNOW 425 W BONITA AVE STE 208 SAN DIMAS CA 91773

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	OLIN CORPORATION FLINCHBAUGH DIV E STEINER B STEWART PO BOX 127 RED LION PA 17356	1	BOEING R BOHLMANN PO BOX 516 MC 5021322 ST LOUIS MO 63166-0516
1	OLIN CORPORATION L WHITMORE 10101 9TH ST NORTH ST PETERSBURG FL 33702	2	BOEING DEFENSE AND SPACE GRP W HAMMOND J RUSSELL S 4X55 PO BOX 3707 SEATTLE WA 98124-2207
1	DOW UT S TIDRICK 15 STERLING DR WALLINGFORD CT 06492	2	BOEING ROTORCRAFT P MINGURT P HANDEL 800 B PUTNAM BLVD WALLINGFORD PA 19086
5	SIKORSKY AIRCRAFT G JACARUSO T CARSTENSAN B KAY S GARBO M S S330A J ADELMANN 6900 MAIN ST PO BOX 9729 STRATFORD CT 06497-9729	1	BOEING DOUGLAS PRODUCTS DIV L J HART SMITH 3855 LAKEWOOD BLVD D800 0019 LONG BEACH CA 90846-0001
1	PRATT & WHITNEY D HAMBRICK 400 MAIN ST MS 114 37 EAST HARTFORD CT 06108	1	LOCKHEED MARTIN S REEVE 8650 COBB DR D 73 62 MZ 0648 MARIETTA GA 30063-0648
1	AEROSPACE CORP G HAWKINS M4 945 2350 E EL SEGUNDO BLVD EL SEGUNDO CA 90245	1	LOCKHEED MARTIN SKUNK WORKS D FORTNEY 1011 LOCKHEED WAY PALMDALE CA 93599-2502
2	CYTEC FIBERITE M LIN W WEB 1440 N KRAEMER BLVD ANAHEIM CA 92806	1	LOCKHEED MARTIN R FIELDS 1195 IRWIN CT WINTER SPRINGS FL 32708
1	HEXCEL T BITZER 11711 DUBLIN BLVD DUBLIN CA 94568	1	MATERIALS SCIENCES CORP B W ROSEN 500 OFFICE CENTER DR STE 250 FORT WASHINGTON PA 19034

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	NORTHROP GRUMMAN CORP ELECTRONIC SENSORS & SYSTEMS DIV E SCHOCH 1745A WEST NURSERY RD MAILSTOP V 16 LINTHICUM MD 21090
2	NORTHROP GRUMMAN ENVIRONMENTAL PROGRAMS R OSTERMAN A YEN 8900 E WASHINGTON BLVD PICO RIVERA CA 90660
1	UNITED DEFENSE LP PO BOX 359 D MARTIN SANTA CLARA CA 95052
1	UNITED DEFENSE LP PO BOX 58123 G THOMAS SANTA CLARA CA 95052
2	UNITED DEFENSE LP MAIL DROP M53 R BARRETT V HORVATIC 328 W BROKAW RD SANTA CLARA CA 95052-0359
3	UNITED DEFENSE LP GROUND SYSTEMS DIVISION M PEDRAZZI MAIL DROP N09 A LEE MAIL DROP N11 M MACLEAN MAIL DROP N06 1205 COLEMAN AVE SANTA CLARA CA 95052
4	UNITED DEFENSE LP 4800 EAST RIVER RD R BRYNSVOLD P JANKE MS170 T GIOVANETTI MS236 B VAN WYK MS389 MINNEAPOLIS MN 55421-1498

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
2	GENERAL DYNAMICS LAND SYSTEMS D REES M PASIK PO BOX 2074 WARREN MI 48090-2074
1	GENERAL DYNAMICS LAND SYSTEMS DIVISION D BARTLE PO BOX 1901 WARREN MI 48090
1	GENERAL DYNAMICS LAND SYSTEMS MUSKEGON OPERATIONS W SOMMERS JR 76 GETTY ST MUSKEGON MI 49442
1	GENERAL DYNAMICS AMPHIBIOUS SYS SURVIVABILITY LEAD G WALKER 991 ANNAPOLIS WAY WOODBIDGE VA 22191
5	INSTITUTE FOR ADVANCED TECH T KIEHNE H FAIR P SULLIVAN W REINECKE I MCNAB 4030 2 W BRAKER LN AUSTIN TX 78759
2	CIVIL ENGR RSCH FOUNDATION H BERNSTEIN PRESIDENT R BELLE 1015 15TH ST NW STE 600 WASHINGTON DC 20005
1	ARROW TECH ASSO 1233 SHELBURNE RD STE D 8 SOUTH BURLINGTON VT 05403-7700

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	CONSULTANT R EICHELBERGER 409 W CATHERINE ST BEL AIR MD 21014-3613	1	UNIVERSITY OF UTAH DEPT OF MECH & INDUSTRIAL ENGR S SWANSON SALT LAKE CITY UT 84112
1	UCLA MANE DEPT ENGR IV H THOMAS HAHN LOS ANGELES CA 90024-1597	2	PENNSYLVANIA STATE UNIV R MCNITT C BAKIS 227 HAMMOND BLDG UNIVERSITY PARK PA 16802
2	U OF DAYTON RESEARCH INSTUTE RAN Y KIM AJIT K ROY 300 COLLEGE PARK AVE DAYTON OH 45469-0168	1	PENNSYLVANIA STATE UNIV RENATA S ENGEL 245 HAMMOND BLDG UNIVERSITY PARK PA 16801
1	MIT P LAGACE 77 MASS AVE CAMBRIDGE MA 01887	1	PURDUE UNIVERSITY SCHOOL OF AERO & ASTRO C T SUN W LAFAYETTE IN 47907-1282
1	IIT RESEARCH CENTER D ROSE 201 MILL ST ROME NY 13440-6916	1	STANFORD UNIVERSITY DEPARTMENT OF AERONAUTICS AND AEROBALLISTICS DURANT BUILDING S TSAI STANFORD CA 94305
1	GEORGIA TECH RESEARCH INSTITUTE GEORGIA INSTITUTE OF TECHNOLOGY P FRIEDERICH ATLANTA GA 30392	1	UNIVERSITY OF DAYTON J M WHITNEY COLLEGE PARK AVE DAYTON OH 45469-0240
1	MICHIGAN ST UNIVERSITY R AVERILL 3515 EB MSM DEPT EAST LANSING MI 48824-1226	7	UNIVERSITY OF DELAWARE CTR FOR COMPOSITE MATERIALS J GILLESPIE M SANTARE G PALMESE S YARLAGADDA S ADVANI D HEIDER D KUKICH 201 SPENCER LABORATORY NEWARK DE 19716
1	UNIVERSITY OF KENTUCKY LYNN PENN 763 ANDERSON HALL LEXINGTON KY 40506-0046		
1	UNIVERSITY OF WYOMING D ADAMS PO BOX 3295 LARAMIE WY 82071		

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
1	UNIVERSITY OF ILLINOIS AT URBANA CHAMPAIGN NATL CENTER FOR COMPOSITE MATERIALS RESEARCH 216 TALBOT LABORATORY J ECONOMY 104 S WRIGHT ST URBANA IL 61801		<u>ABERDEEN PROVING GROUND</u>
		1	COMMANDER US ARMY MATERIEL SYS ANALYSIS P DIETZ 392 HOPKINS RD AMXSY TD APG MD 21005-5071
3	THE UNIVERSITY OF TEXAS AT AUSTIN CENTER FOR ELECTROMECHANICS J PRICE A WALLS J KITZMILLER 10100 BURNET RD AUSTIN TX 78758-4497	1	DIRECTOR US ARMY RESEARCH LAB AMSRL OP AP L APG MD 21005 5066
		115	DIR USARL AMSRL CI AMSRL CI H W STUREK AMSRL CI S A MARK AMSRL CS IO FI M ADAMSON AMSRL SL B J SMITH AMSRL SL BA AMSRL SL BL D BELY R HENRY AMSRL SL BG A YOUNG AMSRL SL I AMSRL WM B A HORST E SCHMIDT AMSRL WM BA W D AMICO F BRANDON AMSRL WM BC P PLOSTINS D LYON J NEWILL S WILKERSON A ZIELINSKI AMSRL WM BD B FORCH R FIFER R PESCE RODRIGUEZ B RICE
3	VA POLYTECHNICAL INSTITUTE & STATE UNIVERSITY DEPT OF ESM M W HYER K REIFSNIDER R JONES BLACKSBURG VA 24061-0219		
1	NORTH CAROLINA STATE UNIVERSITY CIVIL ENGINEERING DEPT W RASDORF PO BOX 7908 RALEIGH NC 27696-7908		
1	UNIVERSITY OF MARYLAND DEPT OF AEROSPACE ENGINEERING ANTHONY J VIZZINI COLLEGE PARK MD 20742		
1	DREXEL UNIVERSITY ALBERT S D WANG 32ND AND CHESTNUT STREETS PHILADELPHIA PA 19104		
1	SOUTHWEST RSCH INSTITUTE ENGR & MATL SCIENCES DIV J RIEGEL 6220 CULEBRA RD PO DRAWER 28510 SAN ANTONIO TX 78228-0510		

NO. OF
COPIES ORGANIZATION

ABERDEEN PROVING GROUND (CONT)

AMSRL WM BE
G WREN
C LEVERITT
D KOOKER
AMSRL WM BR
C SHOEMAKER
J BORNSTEIN
AMSRL WM M
D VIECHNICKI
G HAGNAUER
J MCCAULEY
B TANNER
AMSRL WM MA
R SHUFORD
P TOUCHET
N BECK TAN
D FLANAGAN
L GHORSE
D HARRIS
S MCKNIGHT
P MOY
S NGYUEN
P PATTERSON
G RODRIGUEZ
A TEETS
R YIN
AMSRL WM MB
B FINK
J BENDER
T BLANAS
T BOGETTI
R BOSSOLI
L BURTON
K BOYD
S CORNELISON
P DEHMER
R DOOLEY
W DRYSDALE
G GAZONAS
S GHORSE
D GRANVILLE
D HOPKINS
C HOPPEL
D HENRY
R KASTE
M KLUSEWITZ
M LEADORE
R LIEB

NO. OF
COPIES ORGANIZATION

ABERDEEN PROVING GROUND (CONT)

AMSRL WM MB
E RIGAS
J SANDS
D SPAGNUOLO
W SPURGEON
J TZENG
E WETZEL
A ABRAHAMIAN
M BERMAN
A FRYDMAN
T LI
W MCINTOSH
E SZYMANSKI
AMRSL WM MC
J BEATTY
J SWAB
E CHIN
J MONTGOMERY
A WERESCZCAK
J LASALVIA
J WELLS
AMSRL WM MD
W ROY
S WALSH
AMSRL WM T
B BURNS
AMSRL WM TA
W GILLICH
T HAVEL
J RUNYEON
M BURKINS
E HORWATH
B GOOCH
W BRUCHEY
AMSRL WM TC
R COATES
AMSRL WM TD
A DAS GUPTA
T HADUCH
T MOYNIHAN
F GREGORY
A RAJENDRAN
M RAFTENBERG
M BOTELER
T WEERASOORIYA
D DANDEKAR
A DIETRICH

NO. OF
COPIES ORGANIZATION

ABERDEEN PROVING GROUND (CONT)

AMSRL WM TE
 A NILER
 J POWELL
AMSRL SS SD
 H WALLACE
AMSRL SS SE R
 R CHASE
AMSRL SS SE DS
 R REYZER
 R ATKINSON
AMSRL SE L
 R WEINRAUB
 J DESMOND
 D WOODBURY

<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>	<u>NO. OF</u> <u>COPIES</u>	<u>ORGANIZATION</u>
1	R MARTIN MERL LTD TAMWORTH RD HERTFORD SG13 7DG UK	2	ROYAL MILITARY COLLEGE OF SCIENCE SHRIVENHAM D BULMAN B LAWTON SWINDON WILTS SN6 8LA UNITED KINGDOM
1	PW LAY SMC SCOTLAND DERA ROSYTH ROSYTH ROYAL DOCKYARD DUNFERMLINE FIFE KY 11 2XR UK	1	SWISS FEDERAL ARMAMENTS WKS WALTER LANZ ALLMENDSTRASSE 86 3602 THUN SWITZERLAND
1	T GOTTESMAN CIVIL AVIATION ADMINISTRATION PO BOX 8 BEN GURION INTERNL AIRPORT LOD 70150 ISRAEL	1	PROFESSOR SOL BODNER ISRAEL INSTITUTE OF TECHNOLOGY FACULTY OF MECHANICAL ENGR HAIFA 3200 ISRAEL
1	S ANDRE AEROSPATIALE A BTE CC RTE MD132 316 ROUTE DE BAYONNE TOULOUSE 31060 FRANCE	1	DSTO MATERIALS RSRCH LAB DR NORBERT BURMAN NAVAL PLATFORM VULNERABILITY SHIP STRUCTURES & MATERIALS DIV PO BOX 50 ASCOT VALE VICTORIA AUSTRALIA 3032
1	J BAUER DAIMLER BENZ AEROSPACE D 81663 MUNCHEN MUNICH GERMANY	1	PROFESSOR EDWARD CELENS ECOLE ROYAL MILITAIRE AVE DE LA RENAISSANCE 30 1040 BRUXELLE BELGIQUE
3	DRA FORT HALSTEAD PETER N JONES DAVID SCOTT MIKE HINTON SEVEN OAKS KENT TN 147BP UNITED KINGDOM	1	DEF RES ESTABLISHMENT VALCARTIER ALAIN DUPUIS 2459 BOULEVARD PIE XI NORTH VALCARTIER QUEBEC CANADA PO BOX 8800 COURCELETTE GOA IRO QUEBEC CANADA
1	MR FRANCOIS LESAGE DEFENSE RESEARCH ESTAB VALCARTIER PO BOX 8800 COURCELETTE QUEBEC COA IRO CANADA		

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	INSTITUT FRANCO ALLEMAND DE RECHERCHES DE SAINT LOUIS DE MARC GIRAUD RUE DU GENERAL CASSAGNOU BOITE POSTALE 34 F 68301 SAINT LOUIS CEDEX FRANCE
1	J MANSON ECOLE POLYTECH DMX LTC CH 1015 LAUSANNE SWITZERLAND
1	TNO PRINS MAURITS LAB DR ROB IJSSELSTEIN LANGE KLEIWEG 137 PO BOX 45 2280 AA RIJSWIJK THE NETHERLANDS
2	FOA NAT L DEFENSE RESEARCH ESTAB DR BO JANZON R HOLMLIN DIR DEPT OF WEAPONS & PROTECTION S 172 90 STOCKHOLM SWEDEN
2	DEFENSE TECH & PROC AGENCY GRND MR I CREWTHOR GENERAL HERZOG HAUS 3602 THUN SWITZERLAND
1	MINISTRY OF DEFENCE RAFAEL DR MEIR MAYSELESS ARMAMENT DEVELOPMENT AUTH PO BOX 2250 HAIFA 31021 ISRAEL
1	DR AKE PERSSON DYNAMEC RESEARCH AB PARADISGRND 7 S 151 36 SODERTALJE SWEDEN

<u>NO. OF COPIES</u>	<u>ORGANIZATION</u>
1	ERNST MACH INSTITUT EMI DIRECTOR HAUPTSTRASSE 18 79576 WEIL AM RHEIN GERMANY
1	ERNST MACH INSTITUT EMI DR ALOIS STILP ECKERSTRASSE 4 7800 FREIBURG GERMANY
1	DR IR HANS PASMAN TNO DEFENSE RESEARCH POSTBUS 6006 2600 JA DELFT THE NETHERLANDS
1	DR BITAN HIRSCH TACHKEMONY ST 6 NETAMUA 42611 ISRAEL
1	PROF DR MANFRED HELD DEUTSCHE AEROSPACE AG DYNAMICS SYSTEMS PO BOX 1340 D 86523 SCHROBENHAUSEN GERMANY

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.				
1. AGENCY USE ONLY (Leave blank)		2. REPORT DATE June 2000	3. REPORT TYPE AND DATES COVERED Final, Aug 94-Sep 95	
4. TITLE AND SUBTITLE Thermal Degradation Effects on Consolidation and Bonding in the Thermoplastic Fiber-Placement Process			5. FUNDING NUMBERS AH42	
6. AUTHOR(S) Bruce K. Fink, John W. Gillespie, Jr.,* and Nuri B. Ersoy*				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) U.S. Army Research Laboratory ATTN: AMSRL-WM-MB Aberdeen Proving Ground, MD 21005-5069			8. PERFORMING ORGANIZATION REPORT NUMBER ARL-TR-2238	
9. SPONSORING/MONITORING AGENCY NAMES(S) AND ADDRESS(ES)			10. SPONSORING/MONITORING AGENCY REPORT NUMBER	
11. SUPPLEMENTARY NOTES * University of Delaware, Newark, DE 19716				
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited.			12b. DISTRIBUTION CODE	
13. ABSTRACT (Maximum 200 words) Effects of elevated temperature exposure during thermoplastic fiber placement on bonding and consolidation are investigated experimentally for AS4/polyetherketoneketone (PEKK) composite. Coupons of 24 layers are consolidated on the University of Delaware Center for Composite Materials (UD-CCM) fiber-placement robot at deposition rates of 20 and 40 mm/s over a range of process temperatures (700–900 °C, with 50 °C increments). The main torch and preheater distances and the compaction force are held constant for all coupons. Two competing mechanisms governing strength buildup are considered: (1) polymer bonding and (2) degradation. Coupons are sectioned, and one-half of each coupon is reconsolidated in a hot press at conventional processing conditions (i.e., 30 min at 370 °C and 0.70-MPa pressure) in order to remove any effect of poor consolidation on strength. Void content of the robot-consolidated panels is measured. Short-beam shear (SBS) tests are performed on the specimens cut from each coupon. Strength and void-content measurements for robot-consolidated panels are presented to illustrate the effect of processing parameters on product quality. Results of the SBS strength tests performed on reconsolidated coupons indicate that there is significant decrease in the strength of coupons consolidated at 20-mm/s deposition rate and high torch temperatures, possibly due to polymer degradation, whereas approximately the same value of reconsolidated strength is measured for the 40-mm/s deposition rate, suggesting that polymer degradation is insignificant at that rate. Effects of void content on SBS tests are discussed.				
14. SUBJECT TERMS composites, thermoplastic, tow placement, elevated temperature, degradation			15. NUMBER OF PAGES 58	
			16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT UNCLASSIFIED	18. SECURITY CLASSIFICATION OF THIS PAGE UNCLASSIFIED	19. SECURITY CLASSIFICATION OF ABSTRACT UNCLASSIFIED	20. LIMITATION OF ABSTRACT UL	

INTENTIONALLY LEFT BLANK.

USER EVALUATION SHEET/CHANGE OF ADDRESS

This Laboratory undertakes a continuing effort to improve the quality of the reports it publishes. Your comments/answers to the items/questions below will aid us in our efforts.

1. ARL Report Number/Author ARL-TR-2238 (Fink) Date of Report June 2000

2. Date Report Received _____

3. Does this report satisfy a need? (Comment on purpose, related project, or other area of interest for which the report will be used.) _____

4. Specifically, how is the report being used? (Information source, design data, procedure, source of ideas, etc.) _____

5. Has the information in this report led to any quantitative savings as far as man-hours or dollars saved, operating costs avoided, or efficiencies achieved, etc? If so, please elaborate. _____

6. General Comments. What do you think should be changed to improve future reports? (Indicate changes to organization, technical content, format, etc.) _____

CURRENT
ADDRESS

Organization

Name

E-mail Name

Street or P.O. Box No.

City, State, Zip Code

7. If indicating a Change of Address or Address Correction, please provide the Current or Correct address above and the Old or Incorrect address below.

OLD
ADDRESS

Organization

Name

Street or P.O. Box No.

City, State, Zip Code

(Remove this sheet, fold as indicated, tape closed, and mail.)
(DO NOT STAPLE)