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HIGH GAIN, LOW NOISE AND BROADBAND RAMAN AND BRILLOUIN FIBER-OPTIC AMPLIFIERS, CHANNEL SELECTORS AND SWITCHES

INTERIM TECHNICAL REPORT

CHUNG YU

7/15/93 - 7/14/94

U.S. ARMY RESEARCH OFFICE

DAAL03-91-G-0301

NORTH CAROLINA A&T STATE UNIVERSITY GREENSBORO, NC 27411

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1. BRILLOUIN SCATTERING BASED SENSOR SURVEY

Optical fibers are generally considered to be passive devices; the light transmitted through them at reasonable power levels may be attenuated (or Rayleigh scattered) and reflected, but is not expected to be modulated. Conventional optical fiber sensors utilize the essential passivity of an unperturbed fiber to detect rather small perturbations externally imposed on the fiber.

Conventional passive fiber-is-the-sensor schemes fall into the following categories: interferometric, polarimetric, intensionmetric and modalmetric, utilizing the intrinsic phase, polarization and coherence of light and passive light guiding properties of the fiber, subjected to gross external mechanical perturbations, such as temperature, pressure, strain, corrosion, etc. These perturbations are assumed to translate directly into variations of fiber mechanical properties that affect light beam transmission. Such sensing is gross, and inherently incapable of distinguishing concurrent perturbations, such as temperature and pressure. Various versions of OTDR have also been explored for distributed temperature and strain sensing. However, the sensitivity of these systems is extremely low in standard fibers.

In an active fiber sensor based on stimulated Brillouin scattering, the interaction of light and excited vibrational modes is very different from Brillouin scattering in bulk media. The excitations in a fiber are the acoustic eigenmodes of the cylindrical structure rather than the plane waves of the Brillouin theory for bulk materials, and the acousto-optic interaction takes place in a confined geometry that relaxes the usual Bragg condition. As a result, the scattered light spectrum consists of dozens of lines, each corresponding to a different cylindrical mode rather than the two lines of the plane wave theory.

Stimulated Brillouin scattering (sBs), which is a stimulated backscattering phenomenon, and Guided Acoustic Wave Brillouin Scattering (GAWBS), which is essentially a forward scattering phenomenon, have basically the same stimulated scattering mechanism. The former is the result of strict adherence to the Bragg condition, while the latter is the consequence of relaxation of the Bragg condition. Hence, sBs always exists in any media (bulk or guiding structures, such as the fiber), while GAWBS occurs only in the confined geometry of the fiber.

SBS involves light scattering due to longitudinal phonons, while GAWBS is mainly due to nearresonant transverse acoustic waves in the fiber (see Appendix I).

Recently, sBs based active fiber optic sensors for temperature and strain have been reported while GAWBS fiber based sensing has not been extensively studied. Due to rising interest in this new sensing mechanism, it is timely to review the status of sBs and GAWBS sensor work to date. Such a status review on sBs sensors is given in Table 1. Since sBs theory is well established, only experimental schemes and data are highlighted in the table. We note, in particular, in the work of Bao and Kurashima that sensing fibers are excessively long, and two lasers are used; both these facts run counter to the principle of low cost and system simplicity. In an attempt to achieve single-laser operation, Duffy resorted to the use of polarization-maintaining, birefringent fibers in order to produce two sBs Stokes shifted backscattered light from the fast and slow axes of the fiber. Fiber length required is also reduced due to the small difference in the beat of the Stokes shifts from the two axes, rendering it detectable electronically by high sensitivity heterodyning. However, Duffy's data are still very tentative. We have detected sBs in our experiments and are now exploring the Duffy approach.

It is further noted that sBs temperature and strain sensing is inextricably inter-related, as demonstrated by Bao (Table 1). Thus, there is no means of extracting information on either one, if both are present. Furthermore, if the sBs sensor were to be used as a strain gauge on structures exposed to large ambient temperature fluctuations as in the case of installation on an aircraft, then calibration of the device is almost impossible. It is conceivable that such a device can serve as an ambient temperature sensor, when the fiber is suspended in a strain-free state. This device is thus theoretically single-measurand.

We have studied GAWBS extensively and have always observed the phenomenon at much lower incident laser powers and shorter fiber lengths as compared to the observation of sBs. This immediately connotes the potential superiority of the GAWBS sensor. To explore this fact in depth, the status of the GAWBS theory is revisited with additional confirmation from various viewpoints. A number of concepts are presented in Table 2.

This forward scattering mechanism has apparently been theorized and observed by a number of researchers. Their appearent divergent views are actually consistent when assembled, compared and properly interpreted. For instance, Stone attributed line broadening by Rayleigh-

Brillouin scattering, to diffraction by a finite aperture, which is the radius of the fiber. Based on fiber parameters and laser wavelength, he arrived at a maximum broadening of 500 MHz. Jen analyzed this phenomenon as acousto-optic interaction, arriving at the conclusion that only R_{0m} , R_{2m}, L_{0m}, L_{2m}, TR_{2m}, respectively radial (R), longitudinal (L) and torsional-radial acoustic modes are allowed. This is consistent with Shelby's and Marcuse's findings. Marcuse predicted depolarization of incident laser light in the fiber due to birefringence caused by periodic variations in the fiber diameter. Shelby derived formulas for the experimentally observed fiber acoustic eigenmodes with specific mode frequencies, which have been confirmed by Shiraki, us and others (see Table 3). The depolarizing TR_{2m} modes have been simulated graphically, showing the greatest amount of depolarizing scattering when the center of the fiber core becomes strongly elliptical (Table 5). We note the near absence of scattering by TR₂₁, TR₂₂, TR₂₄, TR₂₆; and high level of scattering by TR23, TR25, TR27 modes (see Table 4). Shelby also predicted extremely low TR mode scattering efficiency, leading us to call such low level scattering 'noise". However, this readily observed 'noise" over the strongly coherent sBs frequency shift is aptly explained by Corvo, who carried out an extensive spatial-temporal Fourier analysis of forward Brillouin scattering. He concluded theoretically that GAWBS would dominate over sBs for long pulse or cw lasers, and also correctly predicted the frequency shift increase with increasing diffraction angle, as observed by us. According to Table 3, Stone's line broadening formula is applied to Shelby, Shiraki and Yu experimental setups. The computed results are tabulated for comparison in Table 6. Maximum GAWBS bandwidth ranges from 900 MHz to about 400 MHz, agreeing reasonably with the data presented by the various groups.

Dependence of sBs Stokes shift simultaneously on temperature and longitudinal strain is a severe drawback of this sensing process. On the other hand, a simple computation carried out in Table 6 clearly indicates the insensitivity of a GAWBS sensor to temperature and longitudinal strain. It is truly a structural integrity sensor. It has also occurred to us that since sBs involves backscattering of light and GAWBS involves forward scattered light, possibility exists of devising a single fiber system for sensing in both the forward and backward directions: backward sensing for temperature and longitudinal strain, and forward sensing for torsional-radial acoustic vibrations in the fiber, that reflect health of a structure to which the fiber is bonded. A setup for dual sensing is shown in Fig.1. The sBs signal has been clearly observed by us as shown in Fig.2.



Table 1: Available SBS Experimental Data As Sensing Mechanism

Kurashima (1990)	Purpose: For distributed temperature sensors	RF Spectrum Analyzer PD 50% F=120MHz F=120MHz F=120MHz FIBER = 2 FIBER = 2 FIBER = 2 FIBER OVEN CHAMBER REFERENCE FIBER = 1 FIBER = 1 FIBER = 1 FIBER = 1	
	-100 -40 -20 0 20 40 60 TEMPERATURE T ('C)	 Laser: Pump - 1.32µm frequency tunable Nd:YAG laser Probe - 1.32µm semiconductor laser Laser linewidth: Pump - 0.4MHz Probe - 1MHz Fiber length: 1.2km Fiber output: Pump - 1.8mW Probe - 0.3mW 	
(1993)	sensing application using birefringent optical fiber.	Half Wave Plas Prism Polarizer Brage Cell Argen Inn Laser Half Wave Prism Polarizer Half Wave Plase High Speed Deactor Prism Polarizer Half Wave Place Half Wave Plase Beservoric Spectrum Analyzer Digital Oucilloscope	
		Parameter: - Laser: 514nm Lexel 3000 Argon-Ton laser - Laser linewidth: 3MHz - Fiber type: Birefringent fiber - Core diameter: 2 μm - Beat length: 1mm	



Table 2

Tuoro 2	Bragg Condition $f_B=(f_B)_{\pi} \sin \frac{\theta}{2}$			
References	s B s (θ=π±δθ)	GAWBS (θ=0°±δθ)		
Stone J.	Fused quartz: (f _B	$\lambda = 12.6$ GHz at $\lambda = 1.3 \mu m, \delta \theta = \lambda / \pi na \approx 0.08$		
(1983)	<u>Linewidth:</u> $\delta f_B = \frac{1}{4} (f_B)_{\pi} (\delta \theta)^2 << 100 \text{MHz}$	<u>Linewidth:</u> $\delta f_B = \frac{1}{2} (f_B)_{\pi} \delta \theta \cong 500 \text{MHz}$		
Jen C.K. (1989)	$\frac{dA_{i}(z)}{dz} = -jC^{*}(z)A_{d}(z)$ $-\frac{dA_{d}(z)}{dz} = -jC(z)A_{i}(z)$ where A_{i} and A_{d} are amplitudes of incident and diffracted optical waves, respectively. $C = C^{i} + C^{z} + C^{rx} + C^{r\phi} + C^{\phi z}$ $C^{z}, C^{z\phi}, C^{\phi z} << C^{t}, C^{z}, C^{z} \cong 5\% C^{t}$ $C^{t} \text{ for } L_{01} = \text{sBs dominates } (L_{0m}, L_{2m})$	$\begin{array}{ll} C^{t} = b [[(E_{r}^{*} \Delta \varepsilon_{rr} E_{r} + E_{\phi}^{*} \Delta \varepsilon_{\phi\phi} E_{\phi}) r dr d\phi \\ C^{z} = b [[(E_{r}^{*} \Delta \varepsilon_{rr} E_{r} + E_{\phi}^{*} \Delta \varepsilon_{\phi\phi} E_{\phi}) r dr d\phi \\ C^{z} = b [[E_{z}^{*} \Delta \varepsilon_{rz} E_{z} r dr d\phi \\ C^{rz} = b [[E_{z}^{*} \Delta \varepsilon_{rz} E_{r} r dr d\phi \\ C^{r\phi} = b [[E_{z}^{*} \Delta \varepsilon_{r\phi} E_{\phi} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = b [[E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*} \Delta \varepsilon_{\phi} E_{z} r dr d\phi \\ C^{\phi z} = 0] (E_{\phi}^{*$		
Marcuse D. (1991)	Predicted depolarization of incident laser light in the fiber by periodic variation in the fiber diameter causing birefringence in the fiber.	$\begin{aligned} r &= a_o + f \sin 2\phi \\ K_{00} &= -i \frac{fK^2 \gamma^2}{4n_i a_o k \beta^2} F \\ F &= 1 + \left(1 - \frac{\gamma^2}{K^2}\right) \frac{J_o^2(Ka_o)}{J_1^2(Ka_o)} + \frac{(\gamma a_o)^2 J_o^3(Ka_o)}{Ka_o J_1^3(Ka_o)} \\ \frac{f\Delta^{3/2}}{a_o^2} L_p &= \frac{\pi V^3}{\sqrt{2}(Ka)^2(\gamma a)^2 F} = 2.5 \\ L_p \text{ is the length required for the polarization to rotate by 90°} \\ a_o = 5 \mu m, \ \Delta = 0.00137, \ V = 2.4 \\ \text{For } f = 0.05 \mu m (1\% \text{ elliptically}), \ L_p = 25m \end{aligned}$		
Corvo A. (1988)	Concluded theoretically that GAWBS would dominate over sBs for long pulse or cw lasers, and also correctly predicted the frequency shift increase with increasing diffraction angle.			
Shelby R.M. (1985)	First group to predict GAWBS.	$\frac{\text{TR}_{2m} \text{ modes } (\underline{q}_{l/2} \otimes \underline{0})}{\text{Light propagating with linear polarization at } \phi = 45^{\circ} \text{ encounter anisotropic perturbations which cause unequal phase shifts for components projected on the } \phi = 0^{\circ} \text{ and } \phi = 90^{\circ} \text{ axes. Polarization components are created along } \phi = 45^{\circ}$ that are frequency up shifted and down shifted by the vibrational mode. The fraction depolarized in this way by the index perturbations is $\eta_{f} \frac{n^{6} \omega^{2} kT (A_{1} - \alpha^{2} A_{2})^{2} (P_{11} - P_{12})^{2}}{64\pi c^{2} \rho V_{s}^{2} a^{2} B_{Tm}} \approx 10^{-12} \text{ cm}^{-1} \text{ for } TR_{25} \text{ mode}$ The frequency shift f_{m} of the light wave scattered due to TR_{2m} modes is expressed as $f_{m} = \frac{V_{s} y_{m}}{\pi d}$ where V_{s} denotes the sound velocity of the shear wave and d is the fiber diameter. y_{m} is the eigenvalue related to the frequency f_{m} of the light scattered by TR_{2m} mode and is obtained by solving the following equation, which is determined by boundary conditions for TR_{2m} modes corresponding to zero traction on the fiber surface, $\left \begin{array}{c} \left[3 - \frac{y_{m}^{2}}{2}\right] f_{2}(\alpha y_{m}) & \left[6 - \frac{y_{m}^{2}}{2}\right] f_{2}(y_{m}) - 3y_{m} f_{3}(y_{m})} \right _{J_{2}(\alpha y_{m}) - \alpha y_{m} f_{3}(\alpha y_{m})} & \left[2 - \frac{y_{m}^{2}}{2}\right] f_{2}(y_{m}) + y_{m} f_{3}(y_{m})} \right _{J_{2}(\alpha y_{m}) - \alpha y_{m} f_{3}(\alpha y_{m})} & \left[2 - \frac{y_{m}^{2}}{2}\right] f_{2}(y_{m}) + y_{m} f_{3}(y_{m})} \right _{J_{2}(\alpha y_{m}) - \alpha y_{m} f_{3}(\alpha y_{m})} & \left[2 - \frac{y_{m}^{2}}{2}\right] f_{2}(y_{m}) + y_{m} f_{3}(y_{m})} \right _{J_{2}(\alpha y_{m}) - \alpha y_{m} f_{3}(\alpha y_{m})} & \left[2 - \frac{y_{m}^{2}}{2}\right] f_{2}(y_{m}) + y_{m} f_{3}(y_{m})} \right _{J_{2}(\alpha y_{m}) - \alpha y_{m} f_{3}(\alpha y_{m})} & \left[2 - \frac{y_{m}^{2}}{2}\right] f_{2}(y_{m}) + y_{m} f_{3}(y_{m})} \right _{J_{3}(x_{m})} & \left[2 - \frac{y_{m}^{2}}{2}\right] f_{3}(y_{m}) + y_{m} f_{3}(y_{m})} \right]_{J_{3}(x_{m})}$		



Table 3 Available GAWBS experimental data



Table 4TR2m Mode Intensity Chart (Yu)

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Mode	Birefringence	Arrow Plot
TR ₂₁	HiBi	
TR ₂₂	LoBi	
TR ₂₃	HiBi	
TR ₂₄	LoBi	
TR ₂₅	Strong HiBi	

 Table 5
 TR_{2m} Mode Birefringence Chart

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TR_{2m} Mode Birefringence Chart (Cont.)

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Mode	Birefringence	Arrow Plot
TR ₂₆	Zero Bi	
TR ₂₇	Strong Bi	
TR ₂₈	HiBi	
TR ₂₉	HiBi	
TR _{2 10}	HiBi	

Reference	Pump Wavelength	Core radius	$\delta\theta = \lambda/\pi na$ n=1.46	$(f_B)_{\pi}$ (GHz)	Observed $(\delta f_B)_{\pi}$ (MHz)	$(\delta f_B)_{\pi} = (f_B)_{\pi} \times \delta \theta/2$ (MHz)
Stone	1.3 μm	3.5 µm	0.08	12.7		500
Shelby	0.647 µm	2.0 µm	0.0686	26.5	800	910
Shiraki	1.55 μm	2.5 μm	0.135	11.3		764
Yu	1.319 µm	4.9 μm	0.0587	13.2	400	387
Perlmutter	0.647 µm	2.5 μm	0.0564	26.5	900	748
Poustie	1.321 μm				875	



* Index of refraction varies widely, so that $(f_B)_{\pi}$ is only approximate.



Fig.1 Setup for Dual Sensing



No fiber spectrum of pump cross talk

1000 m fiber sBs signal

3000 m fiber sBs signal

Laser pump = 72.5 mW

Fig.2 SBS signal from different fiber lengths

2. GAWBS BASED FIBER SENSOR PRELIMINARY TESTS

1 km of single-mode fiber was bonded to a plexiglass plate, which was then mounted on a vibration table as shown in Table 7 (a). Vibration frequencies ranges from 0 - 5000 Hz, and vibration amplitude could also be varied.

In part (b) of Table 7, four holes were drilled in the plexiglass plate. The plate was then mounted and vibrated in a range of frequencies. At 100 Hz, the GAWBS spectrum was maximally distorted, and we identified this as the resonance frequency of the plate with 4 drilled holes.

Four additional holes were drilled as shown in part (c) of Table 7. The GAWBS spectrum was again mounted as the vibration frequency was swept. Resonant frequencies were recorded at 100 Hz, 200 Hz and 300 Hz.

It is clear from these experiments that the GAWBS spectrum is sensitive to structural variations (cracks or holes).

To verify the prediction of Table 8 as to the temperature insensitivity of the GAWBS spectrum, a preliminary experiment was conducted by raising the ambient temperature of the fiber from 0°C to 58°C. The setup is shown in Table 9. A short fiber segment was bonded to a plastic plate and the GAWBS spectrum monitored as the plate was cooled and heated. The GAWBS spectrum did not exhibit noticeable shift or amplitude variations.

One of the key factors in structural degradation is corrosion. Thus, corrosion detection is an essential part of a structural sensor. One set of experiments were carried out by us to ascertain the capability of the GAWBS sensor in sensing chemical composition variation is the fiber cladding. 1 km of pure silica core Sumitomo Z-fiber was connected to 105 m of Sumitomo Gedoped G-fiber. GAWBS spectra of the Z-fiber alone and Z-G combined fiber were recorded and compared. The result is shown in Table 10 (a). For higher order modes, the additional mode peaks due to the short (105 m) G-fiber are clearly visible. Thus, chemical contamination of the fiber, such as that due to corrosion, should be detectable in the GAWBS spectrum.

Shiraki has demonstrated in Table 10 (b), that fiber diameter variations are detectable in the GAWBS spectrum. In his case, two equal length fiber segments (3 km) of different diameter s were used, thus, yielding more peaks of equal strengths. We are proceeding to modify this setup

by having shorter fiber lengths and also of significantly different lengths to observe the detectability of fiber diameter variations by the GAWBS spectrum. These results are expected to translate into sensing application for fiber mechanical perturbations.

Parameters	Experimental Setup	Test Results
(a) 1000 m single- mode fiber test coupon mounted on the vibrating table		
(b) Test jig with 4 drilled holes at 4 corners		GAWBS vibration spectrum. Resonance at 100 Hz
(c) Test jig with 8 drilled holes		GAWBS vibration spectrum at various vibration frequencies. Resonances exhibit at 100, 200, 300 Hz when GAWBS spectra undergo maximum vibrations

Table 7 GAWBS Based Fiber Sensor Jig Mounted on Vibration Table

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Table 8 Calculation of GAWBS Temperature Depende
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Reference	Temperature Range, ΔT	SBS Shift Δ(f _B) _π	Max. GAWBS Shift, $\Delta(\delta f_B)=1/2\Delta(f_B)_{\pi}\delta \theta$		
Bao	$90^{\circ}\text{C} - 10^{\circ}\text{C} = 80^{\circ}\text{C}$	40 MHz	1.6 MHz		
Kurashima	$60^{\circ}\text{C} - (-30^{\circ}\text{C}) = 90^{\circ}\text{C}$	112.5 MHz	4.5 MHz		
Conclusion: GAWBS spectrum is independent of ambient temperature, since max. $\Delta(\delta f_B)$ < 5 MHz, which is not resolvable in the GAWBS spectrum (of 20 MHz scale).					

 Table 9
 Test Jig for Ambient Temperature Variation

Parameters	Experimental Setup	Test Results
Fiber length: 12 m		Mode Amplitude
Laser Power: 145 mW		- No change
Heater: BroilKing potable stove		<u>Mode Frequency</u> - No shift

Reference	GAWBS	Experimental Setup
(a) Yu (1994)	GAWBS spectrum for fiber B	Wave Polarizar Plat Polarizar Parameter - Laser: 1319 nm Nd: YAG ring laser - Laser linewidth: < 10 kHz - Fiber type : A: "G" fiber from Sumitomo B: "Z" fiber from Sumitomo - Fiber length: A: 105 m B: 1000 m
(b) Shiraki K. (1992)	Starting (A) = Star	GAWBS spectrum shift with fiber diameter. Setup similar to (a). <u>Fiber Parameters</u> - Length: 3000 m - Core: Pure silica - Cladding: 1.08 wt% F-doped silica cladding - Fiber A: Outer diameter = 125 μm - Fiber B: Outer diameter = ≈ 100 μm

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RESULTS OF PRELIMINARY STUDY

- 1. The GAWBS spectrum as reported by various groups has been compiled and repeatedly confirmed by us experimentally;
- 2. The sBs Stokes shift as reported by many groups has been experimentally confirmed by us in varying fiber lengths and incident laser power levels;
- 3. Various theoretical models of the GAWBS phenomenon, have been compiled, compared and reconciled with the basic sBs phenomenon, obeying the same Bragg condition;
- 4. SBS as a temperature and strain sensor has been examined. Its limitation of long fiber length and high laser power requirements is noted. Simultaneous temperature and strain induced sBs shift may render single-measurand sensing impossible;
- 5. The superiority of the GAWBS sensing mechanism in terms of short fiber length, low laser power as compared to those required by sBs has been theoretically confirmed. Its temperature independence is indicated and experimentally confirmed. This feature may enable GAWBS as a purely structural sensor;
- 6. Various singlemode fiber lengths have been bonded to plastic plates to simulate to some extent flexible aircraft structures. These plates were mounted on a variable-frequency vibration jig, and subject to mechanical vibrations of varying amplitudes and frequencies. Resonant frequencies were recorded when the GAWBS spectrum exhibited maximum mode amplitude variations. Additional holes were drilled in the plastic plates to simulate cracks and the same vibration tests repeated. Different resonant frequencies were resonant frequencies were recorded. Such resonances were very sharp, to within 1 Hz;

- 7. The free standing singlemode fiber depolarizes light very readily. However, when bonded to the plastic structure, such depolarization was found to be minimal;
- 8. It appears that the GAWBS spectrum shifts readily with fiber diameter and doping variations. This is an excellent indication of the sensing potential of this spectrum for corrosion and strain in the fiber, and hence those of the structure to which the fiber is bonded.

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The vibrational modes of a cylinder are characterized by a transverse displacement profile and a wave vector in the direction parallel to the cylinder axis. The transverse profile can be classified according to the character of their motion as torsional, radial, longitudinal, flexural, or mixtures of these motions. The modes responsible for forward scattering light in the core are radial modes independent of ϕ (R_{0m}) or mixed torsional-radial modes varying sinusoidally as 2ϕ (TR_{2m}), the latter being doubly degenerate. A nearly infinite number of modes and a similar number of nonzero mode frequencies exists for each value of the axial wave vector q_{ll} . This multiplicity of dispersion curves is the key difference between the bulk sBs scattering case and scattering considered here by guided acoustic-wave in an optical fiber. The frequencies detected correspond to axial wave vectors near $q_{ll}=0$.

Polarized scattering results from the radial dilatation or R_{0m} modes which cause pure phase modulation. To detect this effect, a single-mode fiber must be used as one arm of a Mach-Zehnder interferometer. To detect depolarized scattering which is due to the TR_{2m} modes, elliptically polarized light is coupled into the fiber core under conditions where the polarization of the output is identical to the input as in polarization maintaining fibers. A Glan-Thompson polarizer after the fiber rejects the linear polarization corresponding to the major axis of the input. The much weaker minor-axis polarization is directed onto the photodiode and acts as a local oscillator in heterodyne detection of the depolarized scattering of light initially polarized along the major axis.

In a uniform cylinder, the frequencies of these mode depend only on cylinder radius, a, velocity of the longitudinal (dilation R_{0m}) modes, V_d , velocity of the mixed torsional-radial or transverse (shear, TR_{2m}) modes, V_s , and the wave vector along the cylinder axis. Each mode couples the incident and a forward scattered photon, thus giving rise to a Stokes and an anti-Stokes peak in the light scattering spectrum for each mode.

For polarized guided Brillouin spectrum of R_{0m} modes in a single mode fiber:

Scattering efficiency

$$\eta_F = \frac{n^6 \omega^2 k T (P_{11} + P_{12})^2}{32 \pi c^2 \rho V_d^2 a^2 B_{Rm}}$$

 $B_{Rm} = \int_0^2 J_1^2 (y_m x) x dx$

 $P_{11}=0.121$, $P_{12}=0.270$ are the strain-optic coefficient for fused

quartz

n is index of reflection

For the strong R_{03} mode at 127 MHz, $n_F = 4.4 \times 10^{-12} \text{ cm}^{-1}$

For depolarized guided Brillouin scattering spectrum of TR_{2m} modes in a single-mode fiber:

Scattering efficiency

$$A_{1} = (6 - y_{m}^{2})J_{2} (\alpha y_{m})$$
$$A_{2} = (6 - y^{2}/2)J_{2} (y_{m}) - 3 y_{m} J_{3} (y_{m})$$

 B_{Tm} is analogous to B_{Rm}

$$\eta_F = \frac{n^6 \omega^2 k T (A_1 - \alpha^2 A_2)^2 (P_{11} - P_{12})^2}{64 \pi c^2 \rho V_c^2 a^2 B_{\tau_m}}$$

 $\eta = 1.4 \times 10^{-12} \text{ cm}^{-1} \text{ for } \text{TR}_{25}$

 $\Omega_{\rm m} = (V_{\rm d} y_{\rm m})/a$ gives the frequency *m*th R_{0m} mode $y_{\rm m}$ is the *m*th zero of the Bessel function

For fused silica, $V_s = 3740$ m/sec and $V_d = 5996$ m/sec $\Omega_m = V_s y_m/a$ for TR_{2m} mode

Modes with $q_{//} = 4\pi n/\lambda$ cause backscattering of the incident radiation with a frequency shift of $\Omega_m \approx 4\pi n V_d/\lambda$. The linewidth of the observed backward Brillouin scattering is roughly 145 MHz in a fiber. On the basis of the above analysis, a pattern of splittings similar to that due to forward scattering should be detectable in the backscattering case. This light scattering phenomenon constitutes a thermal-noise source within all optical fibers and might ultimately be significant in communications and sensor applications of these devices.

ACOUSTIC COUPLING TO OPTICAL FIBERS

When a plane acoustic wave is incident on an optical fiber, modeled as an infinite cylinder, the pressure in the interior of the fiber is a function of the ratio of radius to the acoustic wavelength. When this ratio is small the pressure is transmitted to the fiber interior without reflection loss, as predicted below.

For a plane wave of sound incident normally to the axis of a fiber is of radius a, density ρ_{e} , speed of sound c_{e} , and of the form

$$p = p_o e^{i(kx - \omega t)} = p_o e^{i(k\gamma \cos \phi - \omega t)}$$

where p_o is the amplitude of the pressure disturbance, ω is the sound frequency, k is the wave number, γ is the radial vector in cylindrical coordinates, and ϕ is the angle between the radial vector and the direction of the incident acoustic wave, the solution is

r < *a* :

$$p = p_o \sum_{m=0}^{\infty} \varepsilon_m i^m B_m \cos m \Phi J_m \left(\frac{c}{c_e} k \gamma\right)$$

r > a:

$$p = p_o \sum \varepsilon_m i^m \cos m \Phi \left[J_m(k\gamma) + D_m H_m^1(K\gamma) \right]$$

Hence $\varepsilon_m = 1$, m=0, $\varepsilon_m - 2$, m>0, ϕ is the azimuthal angle, J_m is the *m*the-order Bessel function and H_m^{-1} is the *m*th-order Hankel function of the first kind. We then find the radial velocity from

$$v = \frac{1}{i\rho\omega} \nabla p$$

and require that the pressure and radial velocity be continuous at $\gamma=a$. After eliminating D_m , we arrive at

$$B_{o} = \frac{1 + \left(\frac{1}{2}ka\right)^{2}}{1 + \left(\frac{1}{2}ka\right)^{2} - 2\left(1 - \frac{\rho c^{2}}{\rho_{e}c_{e}^{2}}\right)\left(\log\frac{1}{2}ka + \gamma\right)\left(\frac{1}{2}ka\right)^{2}}$$

where γ = Euler's constant = 0.577

Since $ka = 2\pi a/\lambda$, where λ is the acoustic wavelength, we have

$$\lim_{a/\lambda\to 0}B_o=$$

For m>0, B_m vanishes at least as fast as B_o and, in addition, is multiplied by the factor $J_m[(c/c_e)k\gamma]$ which, for $\gamma < a$, $c_e > c$ is much less than unity. Thus for acoustic wavelenths long compared to the fiber radius the sound is completely coupled to the fiber interior.