ARMY RESEARCH LABORATORY



An Overview of NDE Methods For Thick Composites And A Proposal For Analysis of Computed Technology Data

Kristen Weight

ARL-TR-516

November 1994



19941208 019

THE STAL THE INSPECTED 1

Approved for public release; distribution unlimited.

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents. Ģ

Citation of manufacturer's or trade names does not constitute an official endorsement or approval of the use thereof.

Destroy this report when it is no longer needed. Do not return it to the originator.

REPORT D	OCUMENTATION P	AGE	Form Approved OMB No. 0704-0188
Public reporting burden for this collection of int gathering and maintaining the data needed, an collection of information, including suggestions Davis Highway. Suite 1204. Artingaton. VA 2220	formation is estimated to average 1 hour per d completing and reviewing the collection of i for reducing this burden, to Washington Hee 24302, and to the Office of Management and	response, including the time for re- information. Send comments rega- idquarters Services, Directorate for Budget, Paperwork Reduction Proj	viewing instructions, searching existing data sources, rding this burden estimate or any other aspect of this r information Operations and Reports, 1215 Jefferson ect (0704-0188), Washington, DC 20503.
1. AGENCY USE ONLY (Leave blar	ok) 2. REPORT DATE November 1994	3. REPORT TYPE AN Final Octo	d dates covered ber 1993 - October 1994
4. TITLE AND SUBTITLE			5. FUNDING NUMBERS
An Overview of NDE M and a Proposal for A	lethods for Thick Comp Inalysis of Computed T	osites omography Data	
6. AUTHOR(S)			•
Kristen Weight			8 DEPEORMING ORGANIZATION
Auro Deserved Labors	AME(S) AND ADDRESS(ES)		REPORT NUMBER
Watertown, MA 02172	2-0001		
ATTN: AMSRL-MA-PC			ARL-TR-516
9. SPONSORING/MONITORING AGI	ENCY NAME(S) AND ADDRESS(ES)	10. SPONSORING/MONITORING AGENCY REPORT NUMBER
11. SUPPLEMENTARY NOTES			
12a. DISTRIBUTION / AVAILABILITY	STATEMENT		12b. DISTRIBUTION CODE
Approved for public	release; distribution	unlimited.	
13. Non-Destructive Evaluation ing service induced dama	ion (NDE) techniques are to ge, dimensioning of object to materials inspection and	used for detecting m s and characterizing	anufacturing flaws, detect- materials. Many methods
damage due to the com	plexity of the material. T	his report will revie	w many of the techniques
currently being used for	the investigation of compo	site materials. The i	importance of determining
the damage modes is crit	ical to the effort for predic	ting the residual str	ength in thick composites
		-	cafea in anex composites.
The study of composite n	naterials is a multidisciplina	ary field, encompassi	ng the study of mechanics,
The study of composite n materials and NDE in or	naterials is a multidiscipling der to effectively understa	ary field, encompassi and the damage accu	ng the study of mechanics, imulation process and the
The study of composite n materials and NDE in or resulting behavior of com This moort will discu	naterials is a multidiscipling rder to effectively understan posite structures.	ary field, encompassi and the damage accu	ng the study of mechanics, imulation process and the
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Top	naterials is a multidiscipling order to effectively understan posite structures. as some of the information mography (CT) images rep	ary field, encompassi and the damage accu which can be gain resenting the damag	ng the study of mechanics, imulation process and the ed from further study and e modes present in impact
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Tor damaged thick composite	naterials is a multidisciplina order to effectively understan posite structures. ass some of the information mography (CT) images rep s, in particular, woven rovi	ary field, encompassi and the damage accu which can be gaine resenting the damag ng S-2 glass panels.	ng the study of mechanics, imulation process and the ed from further study and e modes present in impact A proposal for automating
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Tor damaged thick composite the calculation of the vo	naterials is a multidisciplina order to effectively understan posite structures. ass some of the information mography (CT) images rep s, in particular, woven rovi dume of damage in the co	ary field, encompassi and the damage accu n which can be gaine resenting the damag ng S-2 glass panels. mposite panels will	ng the study of mechanics, imulation process and the ed from further study and be modes present in impact A proposal for automating be presented as well as a
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Tor damaged thick composite the calculation of the vo proposal for separating th	naterials is a multidisciplinate order to effectively understand posite structures. ass some of the information mography (CT) images rep s, in particular, woven rovi dume of damage in the co he different damage modes	ary field, encompassi and the damage accu n which can be gaine resenting the damag ng S-2 glass panels. mposite panels will corresponding to da	ng the study of mechanics, imulation process and the ed from further study and se modes present in impact A proposal for automating be presented as well as a mage types such as matrix
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Tor damaged thick composite the calculation of the vo proposal for separating th cracking, delaminations a 14. SUBJECT TERMS	naterials is a multidisciplinate order to effectively understand posite structures. ass some of the information mography (CT) images rep s, in particular, woven roviolume of damage in the co he different damage modes and fiber breakage.	ary field, encompassi and the damage accu n which can be gained resenting the damag ng S-2 glass panels. mposite panels will corresponding to da	ng the study of mechanics, imulation process and the ed from further study and e modes present in impact A proposal for automating be presented as well as a mage types such as matrix
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Tor damaged thick composite the calculation of the vo proposal for separating th cracking, delaminations a 14. SUBJECT TERMS Nondestructive Evalu	naterials is a multidisciplinate order to effectively understand posite structures. The some of the information mography (CT) images rep s, in particular, woven rovi alume of damage in the co the different damage modes and fiber breakage.	ary field, encompassi and the damage accu which can be gained resenting the damage ng S-2 glass panels. mposite panels will corresponding to data erials, Computed	ng the study of mechanics, imulation process and the ed from further study and e modes present in impact A proposal for automating be presented as well as a mage types such as matrix 15. NUMBER OF PAGES 13 16. PRICE CODE
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Tor damaged thick composite the calculation of the vo proposal for separating th cracking, delaminations a 14. SUBJECT TERMS Nondestructive Evalu Tomography, Damage	naterials is a multidisciplinaterials is a multidisciplinate order to effectively understand posite structures. This some of the information mography (CT) images rep s, in particular, woven rovi- blume of damage in the co- he different damage modes and fiber breakage.	ary field, encompassi and the damage accu- n which can be gained resenting the damage ng S-2 glass panels. mposite panels will corresponding to date rials, Computed	ng the study of mechanics, imulation process and the ed from further study and e modes present in impact A proposal for automating be presented as well as a mage types such as matrix 15. NUMBER OF PAGES 13 16. PRICE CODE
The study of composite n materials and NDE in or resulting behavior of com This report will discu analysis of Computed Tor damaged thick composite the calculation of the vo proposal for separating th cracking, delaminations a 14. SUBJECT TERMS Nondestructive Evalu Tomography, Damage 17. SECURITY CLASSIFICATION OF REPORT	naterials is a multidisciplinate rder to effectively understand posite structures. uss some of the information mography (CT) images rep s, in particular, woven roviolume of damage in the co he different damage modes and fiber breakage. Nation, Composite Materia 18. SECURITY CLASSIFICATION OF THIS PAGE	ary field, encompassi and the damage accu n which can be gained resenting the damage ng S-2 glass panels. mposite panels will corresponding to dat rials, Computed 19. SECURITY CLASSIFIC OF ABSTRACT	ng the study of mechanics, imulation process and the ed from further study and be modes present in impact A proposal for automating be presented as well as a mage types such as matrix 15. NUMBER OF PAGES 13 16. PRICE CODE CATION 20. LIMITATION OF ABSTRAC

298-102

Introduction 1
Nondestructive Damage Characterization 1
Acoustics/Ultrasonics1
Thermography
Holography and Shearography 3
Radiography 4
Microwave
Computed Tomography 4
A Proposal for Analysis of CT Scans
Description of Data
Automatic Volume Calculation 5
Segmenting of Modes 7
Recommendations
References 10

Contents_

. .

. .

-



Page

Figures

-

• •

. .

•

.

6

¥

		Page
1.	Ultrasonic image of a thick composite panel with simulated damage	. 2
2.	Thermograph of stresses around an open hole in a loaded specimen	3
3.	CT image of damage in a composite laminate	. 5
4.	Bimodal histogram depicting modes in a damaged panel	. 6
5.	Histogram depicting characteristic material signature of an undamaged panel	. 6
6.	Comparison of histograms of an undamaged panel (K) to a damaged panel (P)	. 7
7.	Multimodal histogram of a damaged panel	. 8
8.	CT image corresponding to undamaged mode (material signature)	. 8
9.	a. CT image corresponding to mode 1 in histogram of 6.	
	b. CT image corresponding to mode 2 in histogram of 6	. 9

.

Introduction

Nondestructive evaluation (NDE) techniques which can accurately assess the type and extent of internal damage to thick composites are necessary in order to provide input to residual strength analysis of composite materials. Although there are destructive techniques, such as sectioning and photomicroscopy, which can identify the various types of damage, the residual strength cannot subsequently be experimentally tested.

There are many NDE techniques available today which are capable of detecting various types of damage by measuring temperature changes, changes in signal attenuation or distortions in signal transmission[1,2]. Techniques such as thermography and acoustic impact technique are not effective for internal inspection of thick glass composites but are useful for surface or near surface damage. Acoustic and ultrasonic techniques are able to detect damage but are not able to differentiate between damage modes unless they are known *a priori*. Traditional radiography, acoustic and ultrasonic methods can determine the depth of a flaw, however, damage obscured by other flaws cannot be detected. The images obtained using NDE methods are often not of sufficient quality to distinguish damage types without enhancing the image. Even high resolution, full gray-scale images are usually subject to interpretation and the information in these images may not be fully exploited. Some of the most frequently used methods, along with their strengths and weaknesses, will be discussed in the following section.

Computed tomography (CT) reconstructs a full 3-D image of the inspected object and is therefore able to detect damage through the entire thickness. Research in other methods, such as thermography and ultrasonics, is being conducted to implement 3-D reconstruction [3,4]. CT has been used in the medical field for 30 years and as a result, 3-D reconstruction and image enhancement techniques are more advanced than those corresponding to other NDE techniques. This report will present a proposal for automating a damage volume calculation, thereby eliminating variability due to interpretaion from the outcome. A proposal for separating the different damage modes corresponding to damage types such as matrix cracking and delaminations will also be presented.

Nondestructive Damage Characterization

The following methods are in use or in development for the NDE of composite materials. Each method is described briefly along with its strengths and shortcomings. Concise surveys of current state-of-the-art NDE techniques may be found in References [1-2]. The aim of these techniques is to obtain information about the internal material structure in order to detect and differentiate defects and damage. Among internal structures of interest are manufacturing defects such as voids, porosity, cracks and ply debonding, service induced damage due to environmental effects such as corrosion and moisture, and service loads resulting in delaminations, fiber breakage, matrix failure and general material fracture.

Acoustics/Ultrasonics

Ultrasonic techniques involve introducing high frequency mechanical vibrations or an acoustic pulse into a material. The inspection geometry can be either through-transmission or one-sided pulseecho. Through-transmission inspection measures the signal strength that is transmitted through the material which will show reduced strength where a defect exists. The pulse-echo technique measures the returning signal, which will be reflected back earlier when flaws, such as delaminations,

1

are encountered. Pulse-echo can measure the depth of flaws, however if one flaw is obscured by another, the second flaw cannot be detected.

Acousto-ultrasonics uses a specially formulated acoustic pulse and measures the form of the returning or transmitted wave. These techniques require contact, either direct or by immersion in a coupling fluid, with one or both sides of the object being tested. Laser ultrasonics is based on excitation and detection of sound through the use of laser light and does not require direct contact. Emerging technologies based on laser-diode pumping promise more compact and hence portable laser ultrasonic equipment. Methods of enhancing the images are limited, however, research in this area is continuing and has great potential[1,2,6]. Figure 1 shows a sample ultrasonic image containing a square metal shim positioned at the mid-plane of a 1" thick S-2 glass woven roving laminate.



Figure 1. Ultrasonic image of a thick composite panel with simulated damage.

The standard tap test uses either a coin or a special hammer to detect delaminations or disbonds near the surface. Traditionally, this method has been subject to the interpretation of the individual inspector, but current research is focusing on using an acoustic emission sensor to evaluate the response of the tap in order to obtain objective results [7].

Thermography

Infrared (IR) or thermal imaging relies on the fact that an application of heat to the surface of a material will result in surface temperature changes. If the material is without defect the heat will diffuse uniformly, while defects in the material will show variations as the heat diffuses. Two methods for IR imaging are in use. The one-sided approach, in which the front surface of the material is heated and temperature variations are recorded via an IR imaging system on the same side. The two sided method applies heat to one side and uses IR imaging on the other side, with the two sided method being slightly more sensitive. Thermography works well with complex geometry since contact is not required.

Another type of thermal imaging is vibrothermography, which introduces energy to the specimen and uses an IR system to record the resulting temperature variations. The method of introducing energy limits applications, however it is effective for complex geometries [8]. It is also difficult to find methods for reproducing thermal gradients in composites. Advances are being made in signal analysis tools and systems with greater sensitivity[1,3]. Current sensors can detect temperature changes on the order of $0.001^{\circ}C$. This gives a resolution comparable to a common strain gauge. However, sensitivity decreases rapidly with depth of flaw, and as a result the inspection of thick composites is not very effective. Figure 2 depicts a thermogram showing the stress profile of a loaded metallic specimen with an open hole.



Figure 2. Thermograph of stresses around an open hole in a loaded specimen.

Holography and Shearography

A hologram is formed by the superposition of two wave fronts, produced by an object beam and a reference beam and recorded, usually on photographic film. A fringe pattern is produced when the specimen undergoes a small displacement as a result of stressing it by thermal or mechanical means. A defect-free specimen will produce a smoothly varying fringe pattern, while a defect will cause an anomaly in the fringe pattern. In shearography, only one beam is used and the returning object beam is doubly imaged, with one of the images slightly shifted or 'sheared', rather than using a separate reference beam.

Subsurface defects on the order of 3 mm at a depth of 18 mm can be detected using these

3

methods [10]. Current research is focusing on developing a methodology for detection and characterization of defects using a fringe-spacing analysis method which measures the fringes to determine flaw location and size [1,9,10].

Radiography

Radiography relies on x-rays or other energy sources and images the differential absorption of the energetic particles. Flaws or defects that allow particles to pass, be absorbed or scattered can be imaged. Typical systems have achieved sensitivities of better than 1.0% to 2.0% in thickness or density variations. Radiographic imaging methods include x-ray backscatter which uses the portion of the attenuation due to scattering to determine information about material properties, such as density changes, which indicate defects. Reverse geometry is a real-time radioscopy system, shows improved sensitivity and resolution and may soon be transportable. However, for thick materials, deep flaws are obscured by internal features above them.

.

ŝ

Microwave

Microwave techniques function similarly to radiography. Flaws are detected through the differential absorption or backscattering of the energy applied from a beam of microwave energy. Nonconducting composites are best suited for this method. Recent advances particularly in the development of new equipment in the millimeter domain have made this method very promising, however it may be several years until it is developed sufficiently.[1,11]

Computed Tomography

CT has even better resolution (less than 0.1%) than conventional radiography. CT uses x-rays, or other energy sources, to scan a planar slice through a three dimensional object and employs mathematics to reconstruct the interior plane of the object. In this way a full 3-D image can be built, enabling interior damage to be seen. Other techniques collect data from a full 3-D region and any slice can be selected for viewing on a computer. Supervoltage CT and high-resolution x-ray CT are current topics of research and will offer even more accuracy and detail in the future [12,13]. There are portable CT systems available but they are effective only for relatively small objects. CT is an expensive and time consuming technique, but it is capable of detecting almost all types of defects.

A Proposal for the Analysis of CT Scans

This is a proposal of a procedure to automatically calculate the damage volume in CT scans of impact damaged composite materials. Damage volumes were used in [5] in an attempt to correlate it with residual strength. The volume calculation in Reference [5] was done by applying a median filter to the image to highlight the damage. Then a threshold was applied to the filtered image until it started to break up and then reversed until the the image appeared correct to the operator. A volume was calculated by counting the pixels inside the thresholded area and multiplying by the pixel area and the thickness of the CT slice. This method is subject to operator interpretation and is therefore not rigorous. Automating the procedure will eliminate the subjective aspects of the

volume calculation. Techniques to extract damage modes captured in the CT image are also being proposed. The ability to identify damage modes is essential as input to an analysis of residual strength. The following subsections will describe the available data and proposals for an automatic volume calculation and segmenting of damage modes.

Description of Data

The data consists of CT scans from the study conducted in Reference [5]. The panels used in that study were S-2 glass reinforced plastic (GRP). They measured 20 inches by 20 inches by approximately 1.7 inches. Each panel consisted of 63 to 69 plies of woven roving S-2 glass. The panels were subjected to ballistic impact and then nondestructively evaluated using CT.

Automatic Volume Calculation

Figure 3. shows a CT image of impact damage in a GRP panel. This grayscale image offers an excellent view of the damaged interior. A grayscale CT image can be interpreted as a density map, that is, the x-ray attenuation coefficient correlates directly to the density of the material being scanned. A histogram of the x-ray attenuation coefficient versus frequency of occurrence for a damaged panel is presented in Figure 4. Information from an undamaged panel can be used to determine the amount of damage in a damaged panel.



Figure 3. CT image of damage in a composite laminate.

Damaged areas have different densities than undamaged material and appear as different clusters in a histogram. The histogram in Figure 5 displays the characteristic material signature of an undamaged GRP panel. The remainder of this section will elaborate on a procedure using this information.



Figure 4. Bimodal histogram depicting modes in a damaged panel.



Figure 5. Histogram depicting characteristic material signature of an undamaged panel.

Automating the volume calculation begins by using one of the numerous edge detectors available in image processing, such as contour following, Hough transform and principal curves [14,15]. The damaged area is isolated in the image using one of these edge detectors. Studying the bimodal histogram in Figure 4. reveals that the rightmost mode represents the undamaged material as shown in Figure 5. As an example, in Figure 6, K5 is a slice from the undamaged panel (K), and P19, P20 and P21 (panel P) are CT scans from a panel with very little damage (the damage appears as a spike at the left end of the graph). The mean and standard deviation corresponding to the undamaged portion of panel P are approximately the same as those in the undamaged panel K. Some of the difference between the undamaged modes in histograms can be attributed to variation in the composite material. For the sake of robustness, it would be neccessary to take all the slices from an undamaged panel and obtain an average histogram which will then be subtracted from the histograms of damaged panels. Using statistical techniques, this mode will be removed from a damaged histogram leaving only the information describing the damage. Since the resulting histogram is simply a count of pixels corresponding to the different damage modes, it will be an accurate assessment of the amount of damage. The number of pixels in the damage mode divided by the total number of pixels gives a percent of volume damage for that slice. Once a correct count has been found, the pixels corresponding to damage can be turned on (or off) in the image to provide an approximate view of the damage in the panel.





Segmenting of Modes

ð

)

The techniques proposed in the previous section for removing the undamaged mode will be extended to segment other modes occurring in multimodal CT scans. Figure 7 shows a multimodal histogram corresponding to a CT image of a damaged panel. Figures 8, 9a and 9b show images which represent the individual modes from the histogram in Figure 7. Figure 9a shows an image representing mode 1 and Figure 9b corresponds to mode 2 of the histogram in Figure 7. Figures 9a and 9b demonstrate that the different modes correspond to specific regions which needs to be correlated with specific damage types.



Figure 7. Multimodal histogram of a damaged panel.



Figure 8. CT image corresponding to undamaged mode in a damaged panel.



Figure 9. a. CT image corresponding to damage mode 1 in histogram of 8. b. CT image corresponding to damage mode 2 in histogram of 8.

Recommendations

5

Э

It is recommended that these proposals be carried out as detailed above. The automatic volume calculation is a simple, accurate way to determine the volume of damage in a panel and can be used in additional studies such as those in Reference [5]. The capability of segmenting modes offers great potential benefit as input to residual strength analyses [16]. It is recommended that damaged panels be subjected to photomicrosopy for specific identification of damage types. It is also suggested that cross sectional CT scans be taken, which would more readily identify delaminations, and correlate with the modes in the histograms. The correlation will provide the capability to identify various damage modes, while still retaining the ability to conduct subsequent residual strength tests, and in addition, to assess repair/no-repair options for composites in service applications.

References

- Carriveau, G. W., "Benchmarking of the State-of-the-Art in Nondestructive Testing/Evaluation for Applicability in the Composite Armored Vehicle (CAV) Advanced Technology Demonstrator (ATD) Program," TARDEC TC No. 13604, (November 1993).
- [2] Adams, R. D. and Cawley, P., "A Review of Defect Types and Nondestructive Testing Techniques for Composites and Bonded Joints," NDT International, Vol. 21, No. 4, p. 208-222, (1988).
- [3] Jones, T. and Berger, H., "Thermographic Detection of Impact Damage in Graphite-Epoxy Composites," *Materials Evaluation*, **50(12)**, pp. 1446-1453, (1992).
- [4] Papadakis, E. P., "Future of Ultrasonics", *Materials Evaluation*, pp. 1180-1184, (September 1991).

(

(

l

- [5] Chou, S. C. and DeLuca, E., ed., "Dynamic Response of S-2 Glass Reinforced Plastic Structural Armor: A Progress Report", ARL-SR-5, (December 1993).
- [6] Bar-Cohen, Y., Mal, A. K. and Lih, S., "NDE of Composite Materials Using Ultrasonic Oblique Insonification", *Materials Evaluation*, pp. 1285-1296, (November 1993).
- [7] Raju, P. K., Patel, J. R., and Vaidya, U. K., "Characterization of Defects in Graphite Fiber Based Composite Structures Using the Acoustic Impact Technique (AIT)", Journal of Testing and Evaluation, pp.377-395, (1993).
- [8] Russell, S.S. and Henneke, E. G., II, "Vibrothermographic Inspection of a Glass-Fiber Epoxy Machine Part," *Materials Evaluation*, pp. 870-874, (July 1991).
- [9] Fallstrom K.E., "A Nondestructive Method to Detect Delaminations and Defects in Plates", NDT & E International, pp. 68-76, (April 1991).
- [10] Ravindran, V. R. et al, "Holographic Nondestructive Testing of Low-Modulus Materials", Materials Evaluation, pp. 1058-1065, (September 1992).
- [11] Gopalsami, N., et al, "Millimeter-Wave Imaging for Nondestructive Evaluation of Materials", Materials Evaluation., pp. 412-415, (March 1993).
- [12] Tonner, Paul D., Stanley, J. H., "Supervoltage Computed Tomography for Large Aerospace Structures," *Materials Evaluation*, pp. 1434-1438, (December 1992).
- [13] London, B., Yancey, R. N. and Smith, J. A., "High-Resolution X-Ray Computed Tomography of Composite Materials", *Materials Evaluation*, pp. 604-608, (May 1990).
- [14] Banfield, J. D. and Raftery, A. E., "Ice Floe Identification in Satellite Images Using Mathematical Morphology and Clustering About Principal Curves", Journal of the American Statistical Association, pp.7-16, (1992).
- [15] Ballard, D. H. and Brown, C. M., "Boundary Detection", Computer Vision, Chapter 4, Prentice-Hall, Inc. (1982).
- [16] Saether, E., "Modeling Considerations in the Prediction of Residual Strength in Composite Laminates", U.S. Army Research Laboratory, ARL MR 211, (1994).

DISTRIBUTION LIST

. .

0

)

Э

<u>_</u>`)

lo. of opies	То
1	Office of the Under Secretary of Defense for Research and Engineering, The Pentagon, Washington, DC 20301
1 1 1	Director, U.S. Army Research Laboratory, 2800 Powder Mill Road, Adelphi, MD 20783-119 ATTN: AMSRL-OP-SD-TP, Technical Publishing Branch AMSRL-OP-SD-TA, Records Management AMSRL-OP-SD-TL, Technical Library
2	Commander, Defense Technical Information Center, Cameron Station, Building 5, 5010 Duke Street, Alexandria, VA 23304-6145 ATTN: DTIC-FDAC
1	MIA/CINDAS, Purdue University, 2595 Yeager Road, West Lafayette, IN 47905
1	Commander, Army Research Office, P.O. Box 12211, Research Triangle Park, NC 27709-2211 ATTN: Information Processing Office
· 1	Commander, U.S. Army Materiel Command, 5001 Eisenhower Avenue, Alexandria, VA 22333 ATTN: AMCSCI
1	Commander, U.S. Army Materiel Systems Analysis Activity, Aberdeen Proving Ground, MD 21005 ATTN: AMXSY-MP, H. Cohen
1	Commander, U.S. Army Missile Command, Redstone Arsenal, AL 35809 ATTN: AMSMI-RD-CS-R/Doc
1	Commander, U.S. Army Armament, Munitions and Chemical Command, Dover, NJ 07801 ATTN: Technical Library
1	Commander, U.S. Army Natick Research, Development and Engineering Center Natick, MA 01760-5010 ATTN: SATNC-MI, Technical Library
1	Commander, U.S. Army Satellite Communications Agency, Fort Monmouth, NJ 07703 ATTN: Technical Document Center
1 1	Commander, U.S. Army Tank-Automotive Command, Warren, MI 48397-5000 ATTN: AMSTA-ZSK AMSTA-TSL, Technical Library
1	President, Airborne, Electronics and Special Warfare Board, Fort Bragg, NC 28307 ATTN: Library
1	Director, U.S. Army Research Laboratory, Weapons Technology, Aberdeen Proving Ground, MD 21005-5066 ATTN: AMSRL-WT

No. of	
Copies	То
	10
	Commander Duaway Braving Crowned LIT 04022
-	Commander, bugway Proving Ground, UT 84022
I	ATTN: Technical Library, Technical Information Division
	Commander U.S. Army Research Loboratory 2000 Doubles Mill Day L.A.L.L.L. M.D. Domos
1	ATTNI AMERICA, CO. ATTNI Research Laboratory, 2000 Powder Mill Road, Adeipni, MD 20783
I	ATTN: AMSRL-SS
	Director, Benet Weapons Laboratory LCWSL LISA AMCCOM Watervliet NY 12180
1	ATTN: AMSMC-I CB-TI
1	
	AMSMC-LCB-R
1	AMSMC-LCB-RM
1	AMSMC-LCB-RP
	Commander U.S. Army Fereign Science and T. J. J. Commander Science and T. J. J.
	Charletter, U.S. Army Poreign Science and Technology Center, 220 7th Street, N.E.,
_	chanottesville, VA 22901-5396
3	ATTN: AIFRTC, Applied Technologies Branch, Gerald Schlesinger
	Commander, U.S. Army Aeromedical Research Unit, P.O. Rev 577, Fast Busham, AL, 20200
1	ATTN: Tophing Library
•	ATTN. Technical Library
	U.S. Army Aviation Training Library, Fort Rucker Al. 36360
1	ATTN: Building 5906-5907
•	A HA. Building 5506-5507
	Commander, U.S. Army Agency for Aviation Safety, Fort Busker, AL, 2020
1	ATTN: Technical Library
1	ATTN: Technical Library
	Commander, Clarke Engineer School Library 3202 Nebraska Ave. N. Fort Learnerd Mercu
	MO 65473-5000
1	
	ATTN: LIDrary
	Commander, U.S. Army Engineer Waterways Experiment Station, B.O. Boy 621, Market
	MS 39180
1	ATTN: Research Conter Library
1	ATTIN. Research Center Library
	Commandant, U.S. Army Quartermaster School, Fort Lee, VA 23801
1	ATTN: Quartermaster School Library
·	
	Naval Research Laboratory, Washington, DC 20375
1	ATTN: Code 6384
	Chief of Nevel Breaster A. K. A. State and a
4	ATTN O I VAVAI Research, Arlington, VA 22217
I	ALLN: Code 4/1
	Commander, U.S. Air Force Wright Research and Development Contaction and Development
	Air Force Base, OH 45433-6523
1	ATTN: WRDC/MLLP M Formary In
1	WIDDOWILLE, M. FORNEY, JF.
I	WRDC/MLBC, Mr. Stanley Schulman
	U.S. Department of Commerce, National Institute of Standard and Table 1
	MD 20899
1	

• •

۰.

í

(

£

l

1 ATTN: Stephen M Hsu, Chief, Ceramics Division, Institute for Materials Science and Engineering

No. of Copies	То
1	Committee on Marine Structures, Marine Board, National Research Council, 2101 Constitution Avenue, N.W., Washington, DC 20418
1	Materials Sciences Corporation, Suite 250, 500 Office Center Drive, Fort Washington, PA 19034
1	Charles Stark Draper Laboratory, 555 Technology Square, Cambridge, MA 02139
1	General Dynamics, Convair Aerospace Division, P.O. Box 748, Fort Worth, TX 76101 ATTN: Mfg. Engineering Technical Library
1	Plastics Technical Evaluation Center, PLASTEC, ARDEC, Bldg. 355N, Picatinny Arsenal, NJ 07806-5000 ATTN: Harry Pebly
1	Department of the Army, Aerostructures Directorate, MS-266, U.S. Army Aviation R&T Activity - AVSCOM, Langley Research Center, Hampton, VA 23665-5225
1	NASA - Langley Research Center, Hampton, VA 23665-5255
1	U.S. Army Vehicle Propulsion Directorate, NASA Lewis Research Center, 2100 Brookpark Road, Cleveland, OH 44135-3191 ATTN: AMSRL-VP
1	Director, Defense Intelligence Agency, Washington, DC 20340-6053 ATTN: ODT-5A, Mr. Frank Jaeger
1	U.S. Army Communications and Electronics Command, Fort Monmouth, NJ 07703 ATTN: Technical Library
1	U.S. Army Research Laboratory, Electronic Power Sources Directorate, Fort Monmouth, NJ 07703 ATTN: Technical Library
2 5	Director, U.S. Army Research Laboratory, Watertown, MA 02172-0001 ATTN: AMSRL-OP-WT-IS, Technical Library Author

13

 \rangle