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ARMSTRONG LABORATORY

MODELING RESPIRATORY GAS DYNAMICS IN THE AVIATOR'S BREATHING SYSTEM

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Final Technical Report for Period May 1993 - December 1993

Approved for public release; distribution is unlimited.

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BROOKS AIR FORCE BASE, TEXAS

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This report has been reviewed and is approved for publication.

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13. ABSTRACT *(Maximum 200 words)*

Biodynamic Research Corporation (BRC) completed an SBIR Phase I project to study the feasibility of developing a model of the Aviator's Breathing System (ABS). The motivation for the project was the desire to develop a model which could simulate the cardiovascular and respiratory responses to altitude and acceleration stress encountered in high performance military aircraft. Software modules were developed and tested for simulation of: (1) the flows and pressures within the breathing gas delivery system; (2) the flows, pressures, and gas distribution within the lung; and (3) the steady-state flows and pressures within the cardiovascular system. Subprograms were also developed to compute altitude barometric pressure relationships as well as passenger cabin pressures in military aircraft. In addition to the software development, BRC reviewed and organized the Government furnished data from a series of manned rapid decompression known as the EONS Experiments. The data from approximately 170 experimental decompressions were screened for their suitability for use in parameter selection and validation of the respiratory modeling software. The data appears to be highly coherent and fully usable for model validation. We conclude that the development of an integrated ABS Model is feasible and desirable.

14. SUBJECT TERMS

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Mathematical model
Rapid decompression
Respiratory physiology

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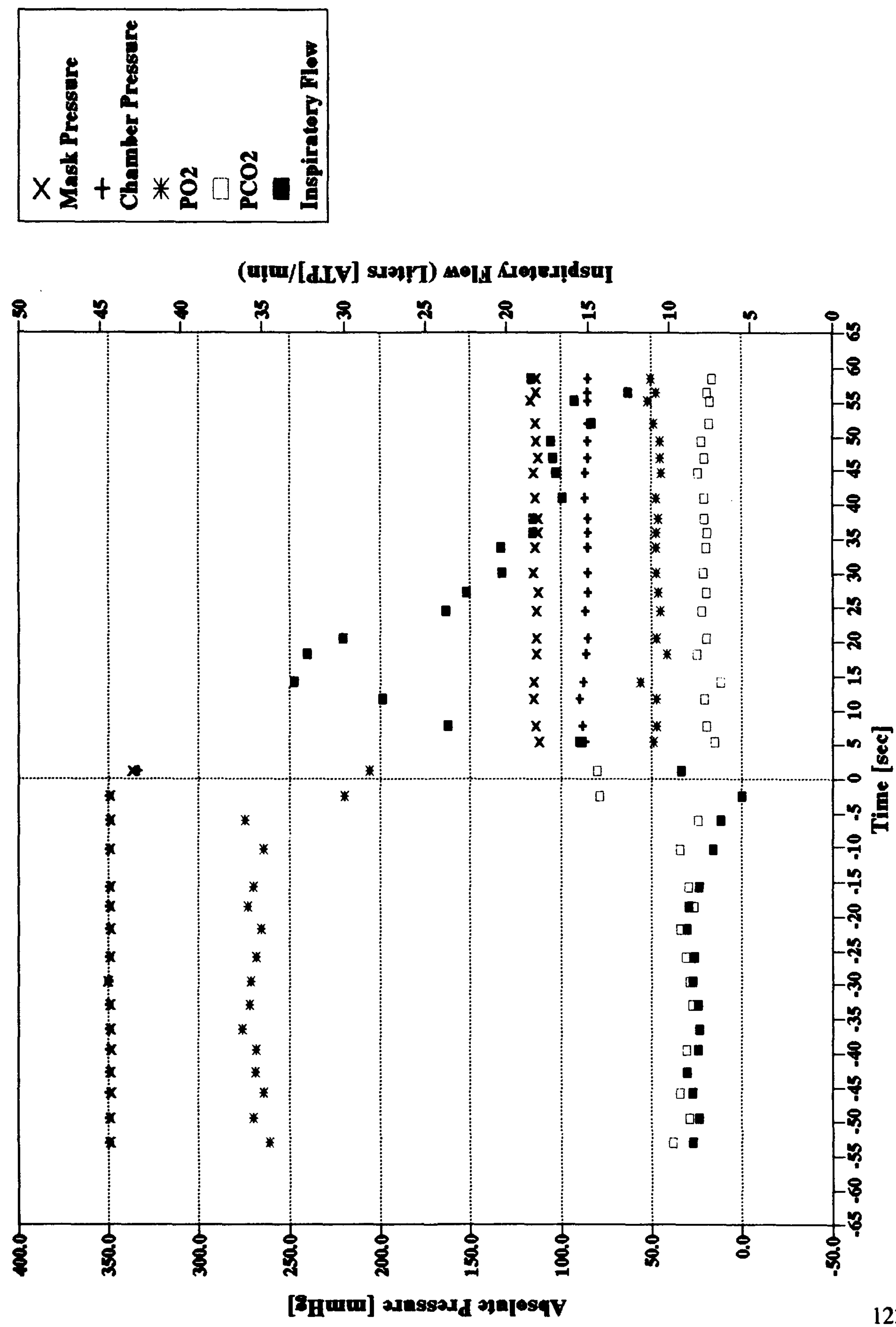
APPENDICES

APPENDIX A

Rapid Decompression Data Plots

Subject: BES / 100% O2 Non-Dilution
20/50 kft Rapid Decompression



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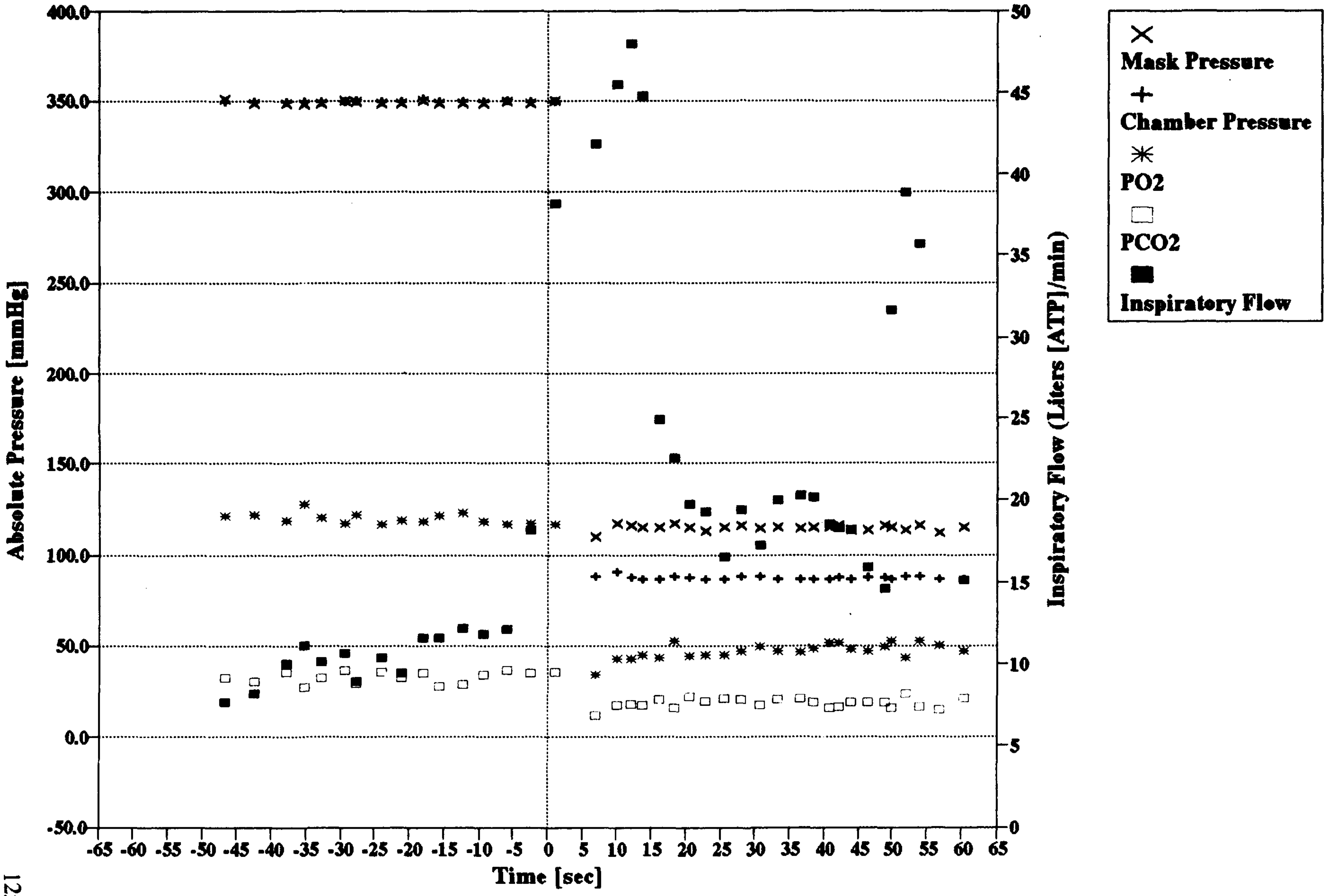

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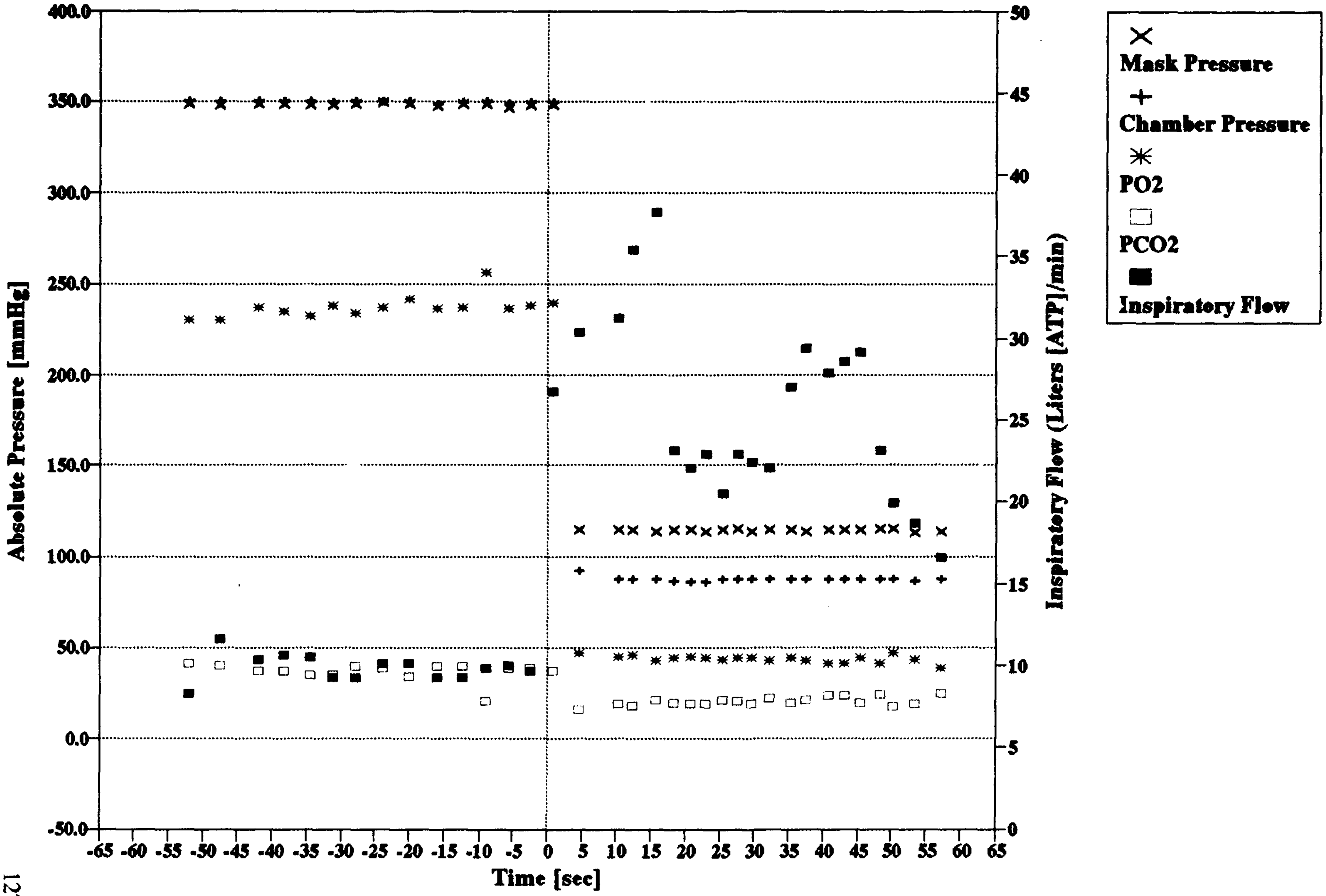

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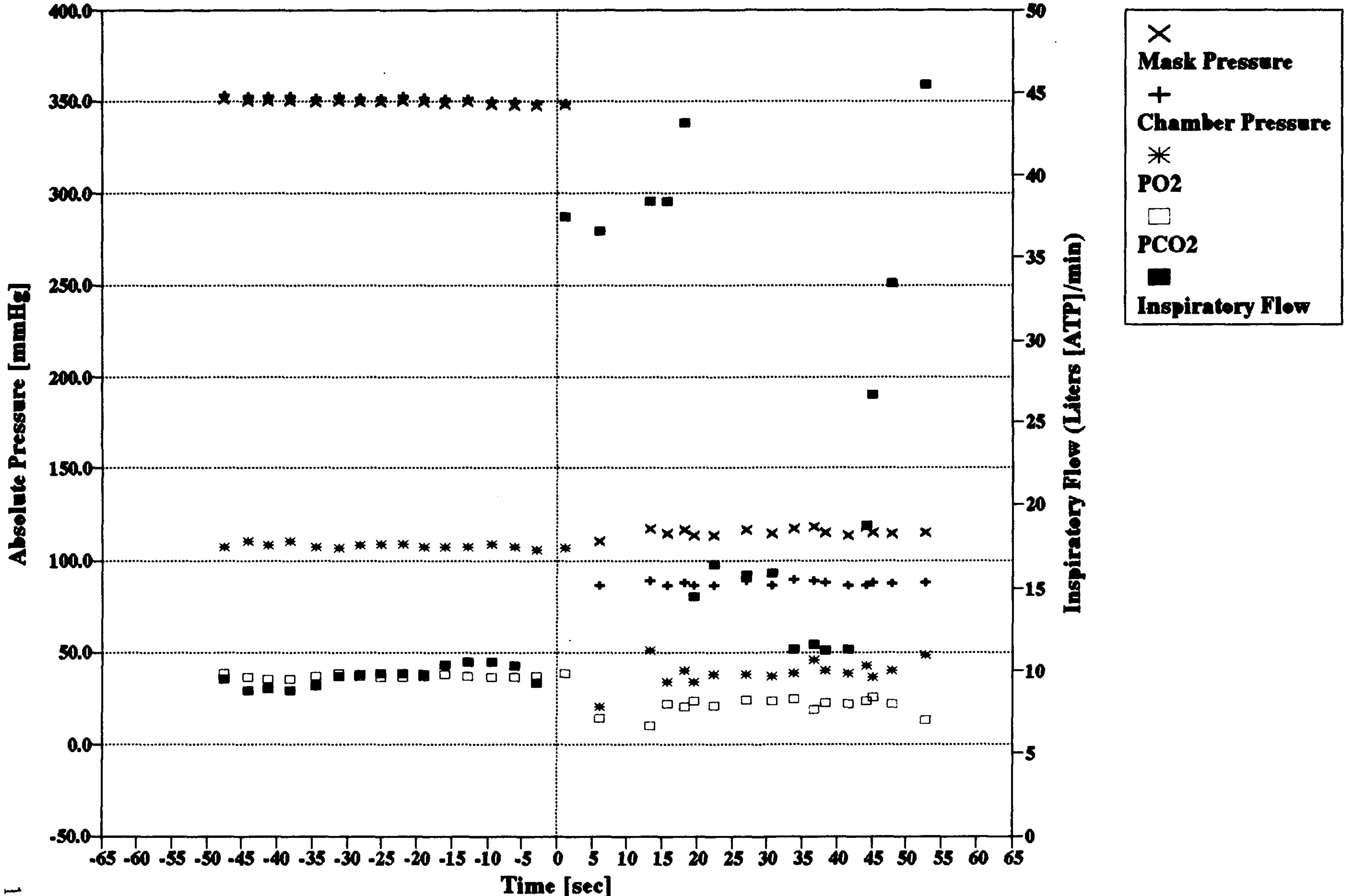

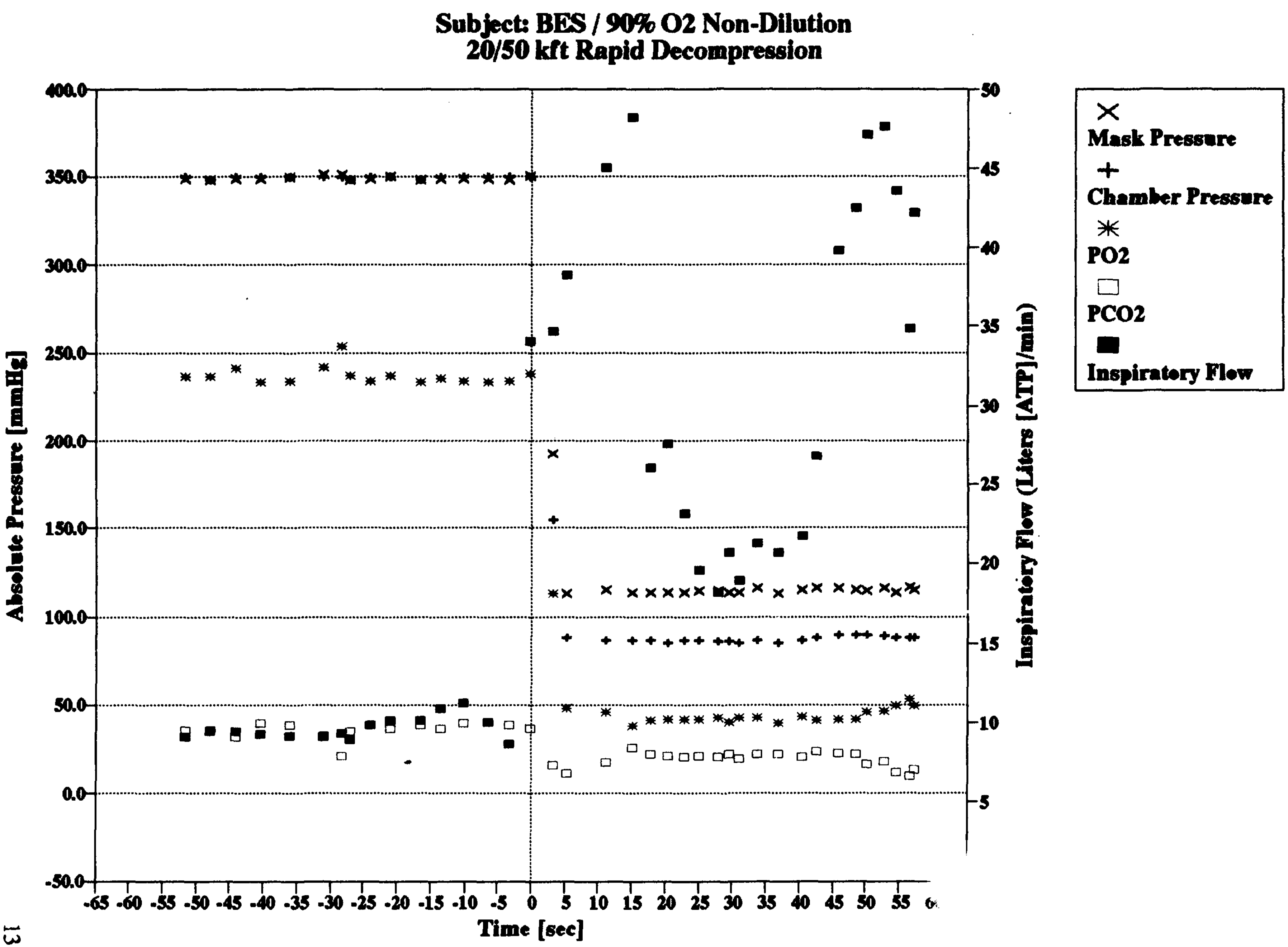

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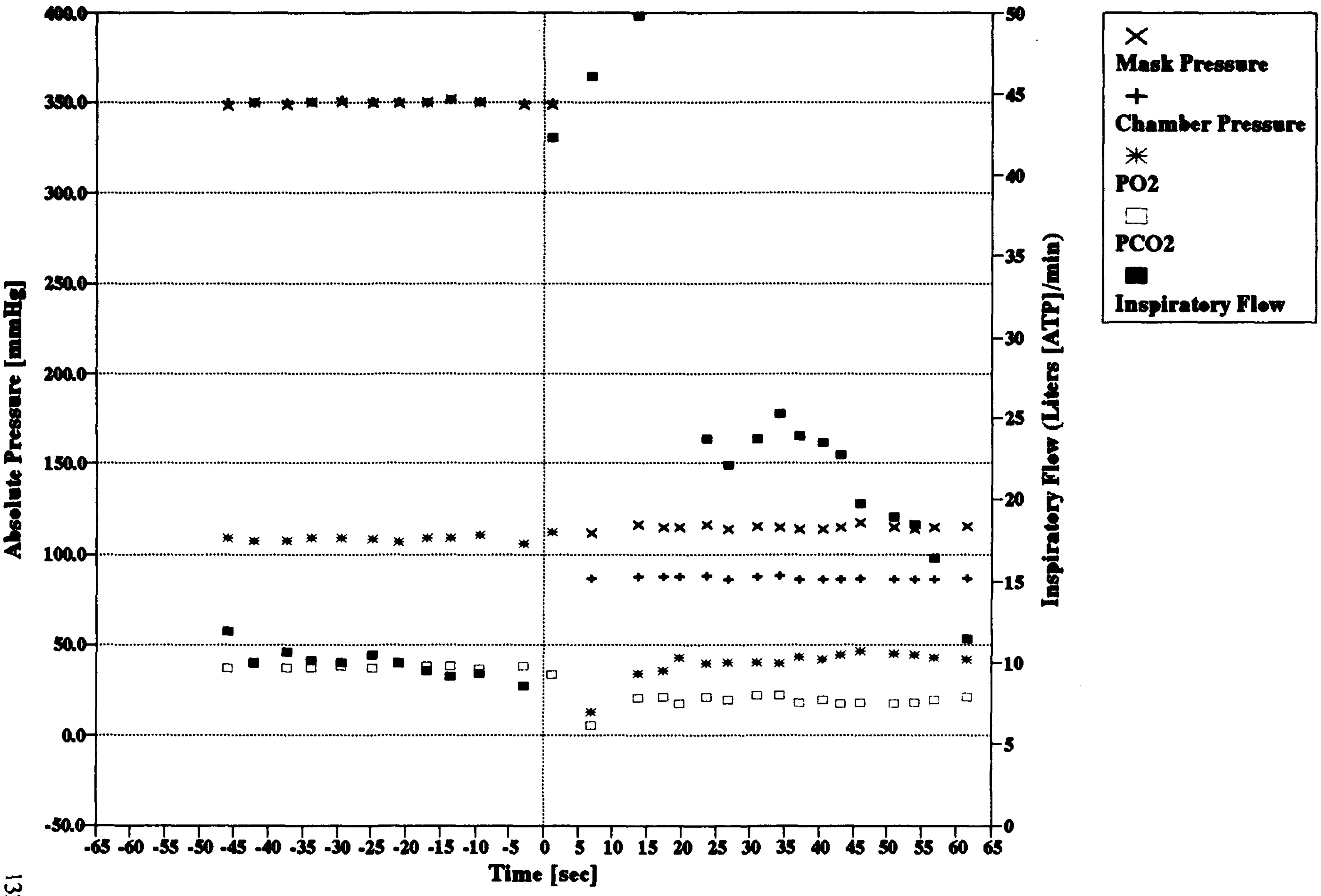
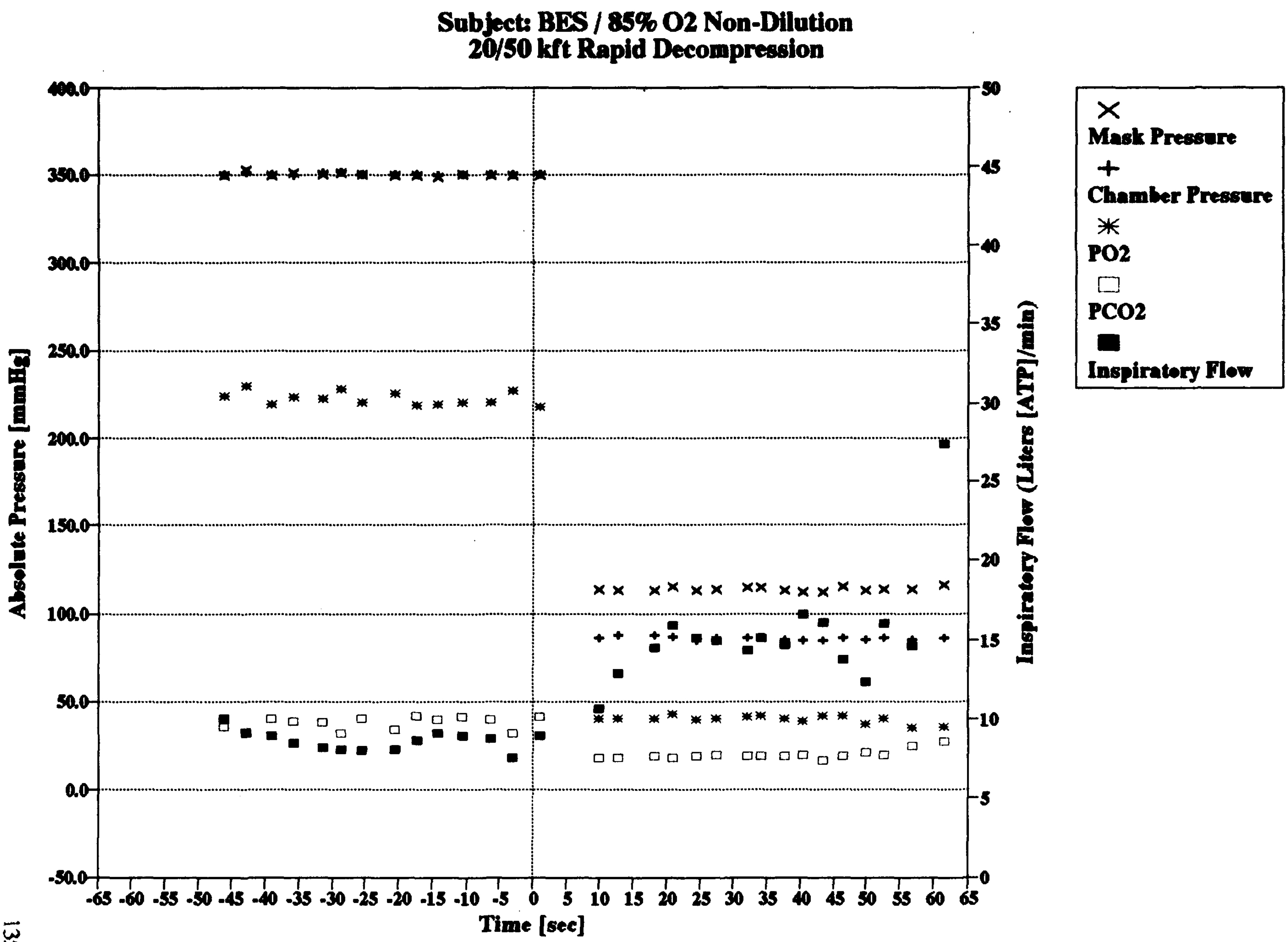
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: BES / 90% O2 Dilution
20/50 kft Rapid Decompression





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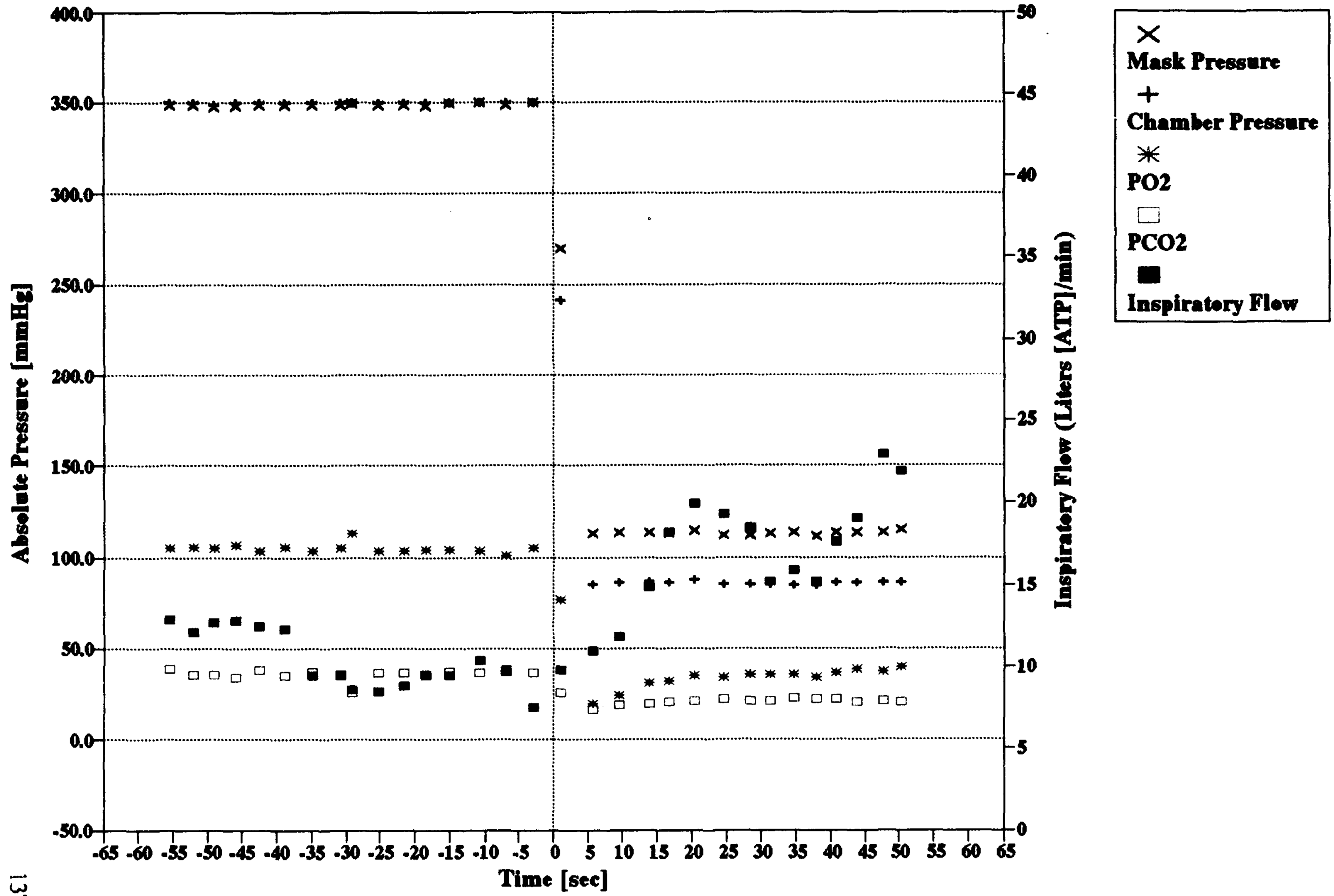
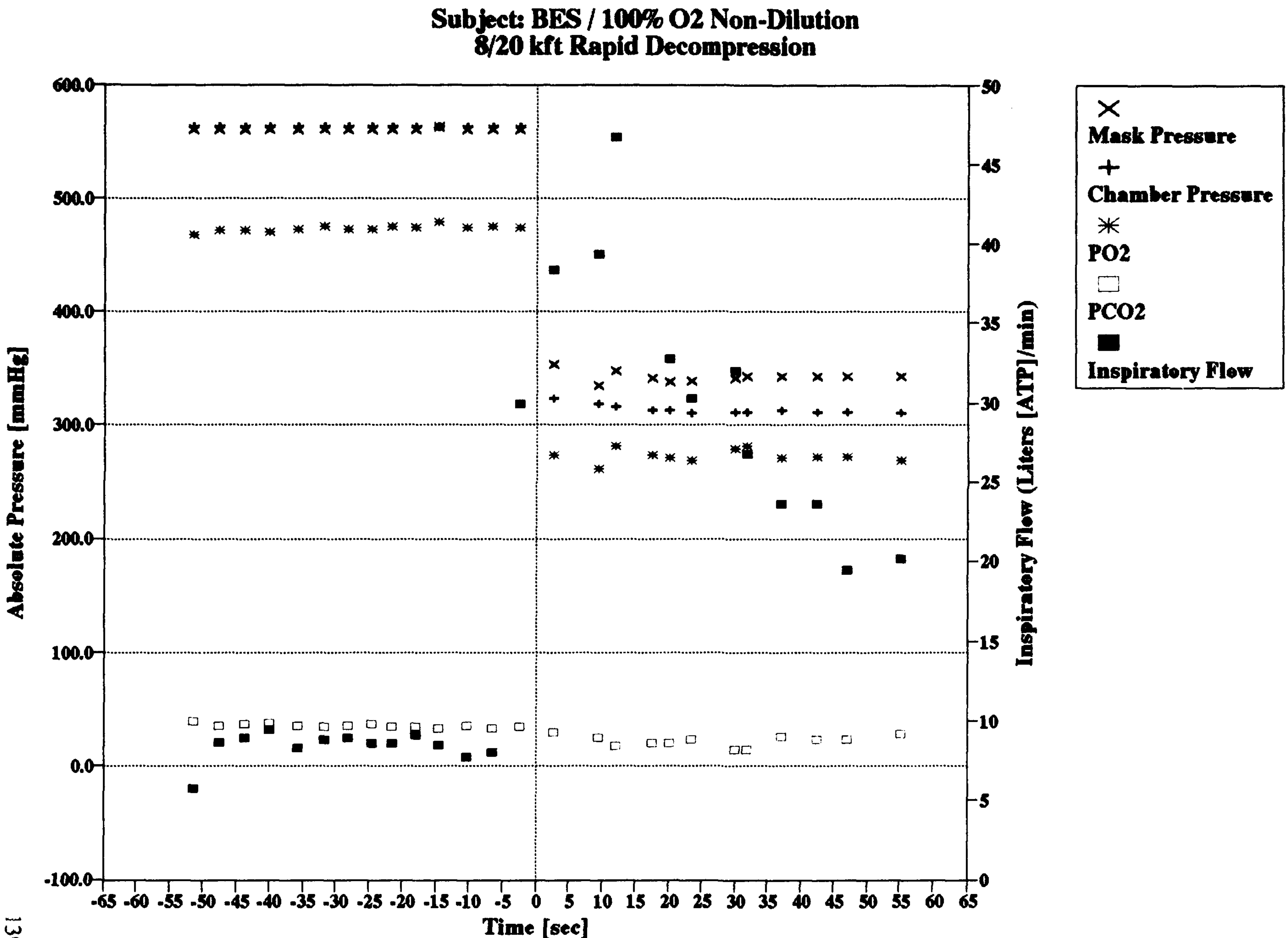
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: BES / 85% O2 Dilution
20/50 kft Rapid Decompression

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Absolute Pressure [mmHg]
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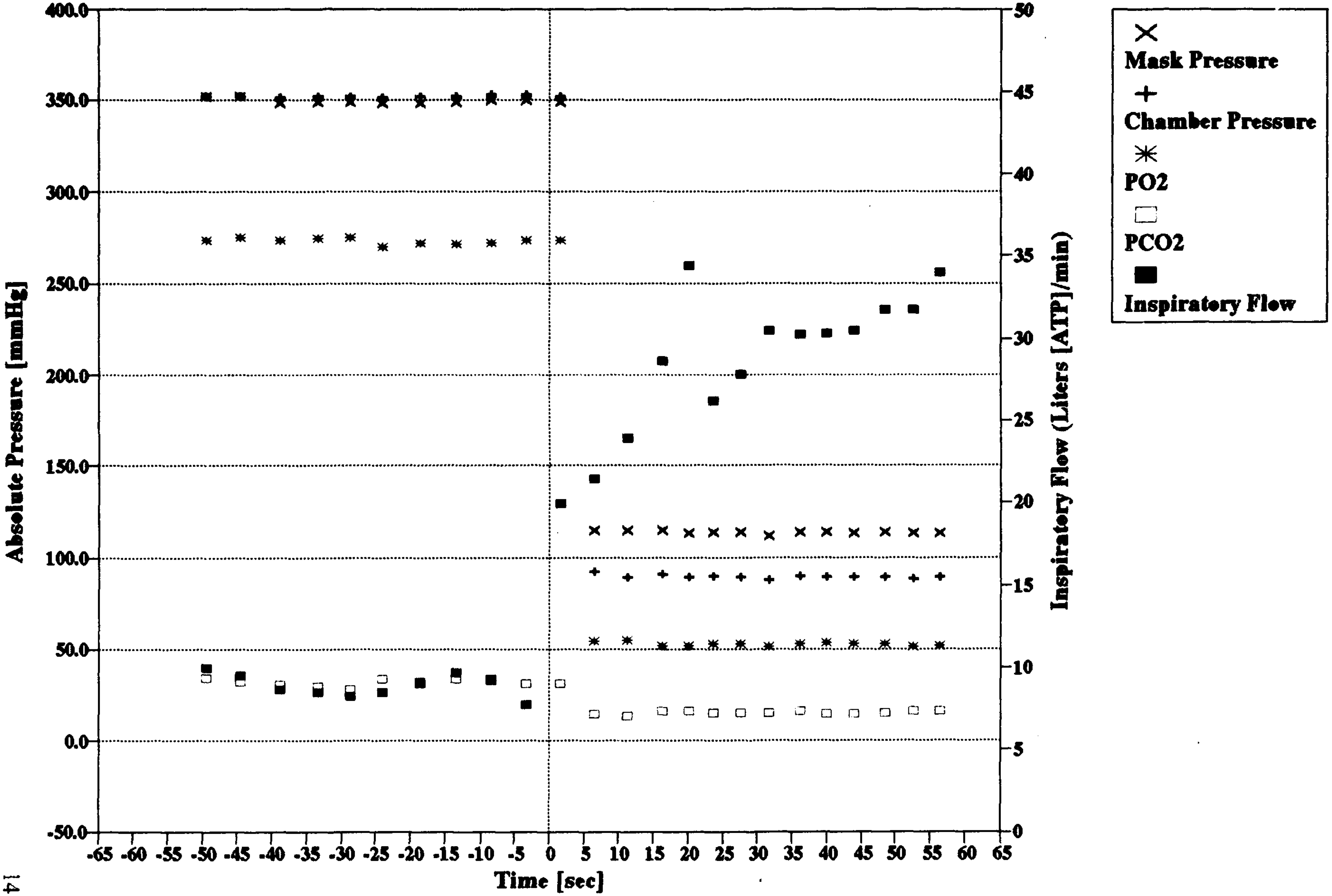
500.0
400.0
300.0
200.0
100.0
0.0
-100.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: BOM / 100% O2 Non-Dilution
20/50 kft Rapid Decompression



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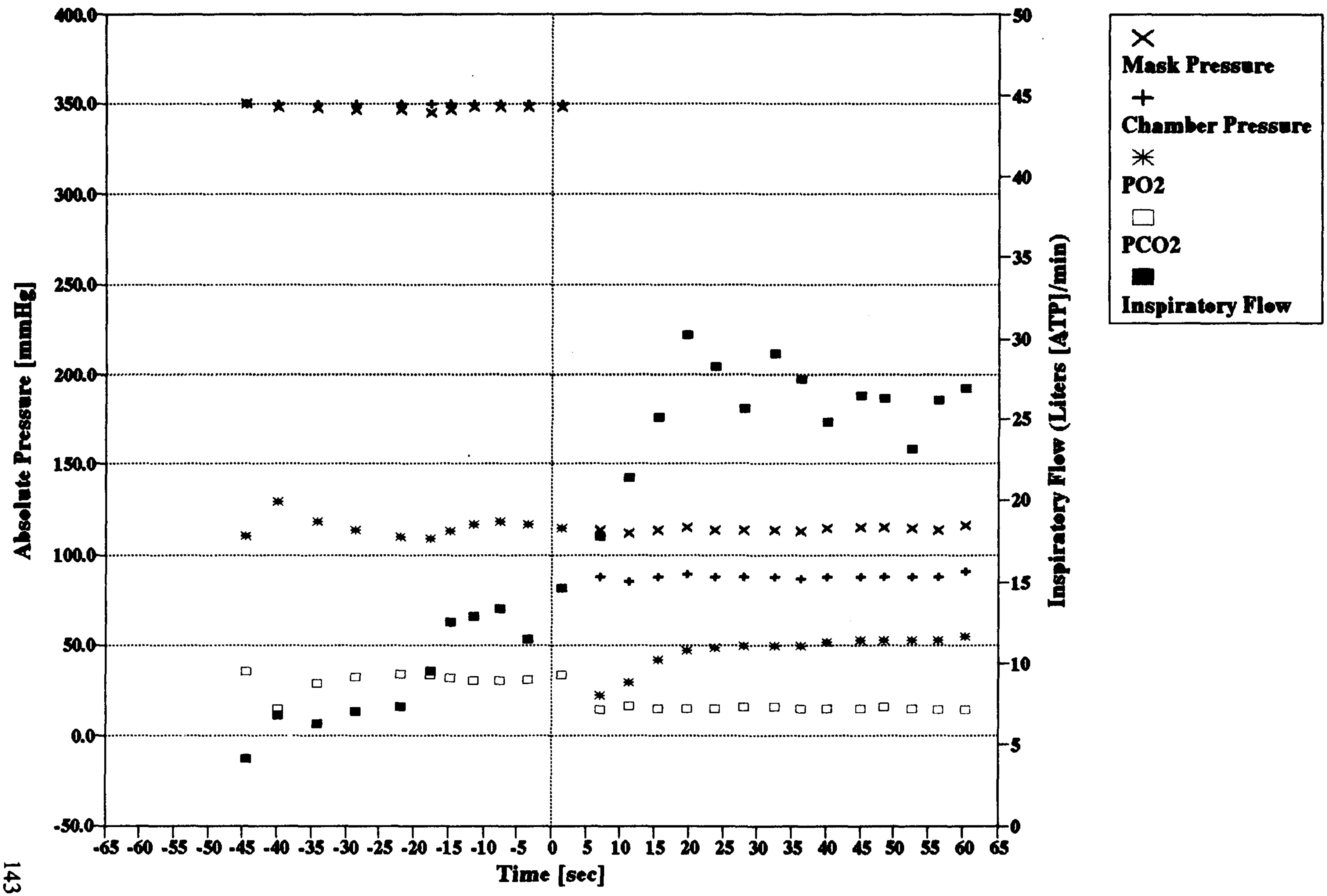

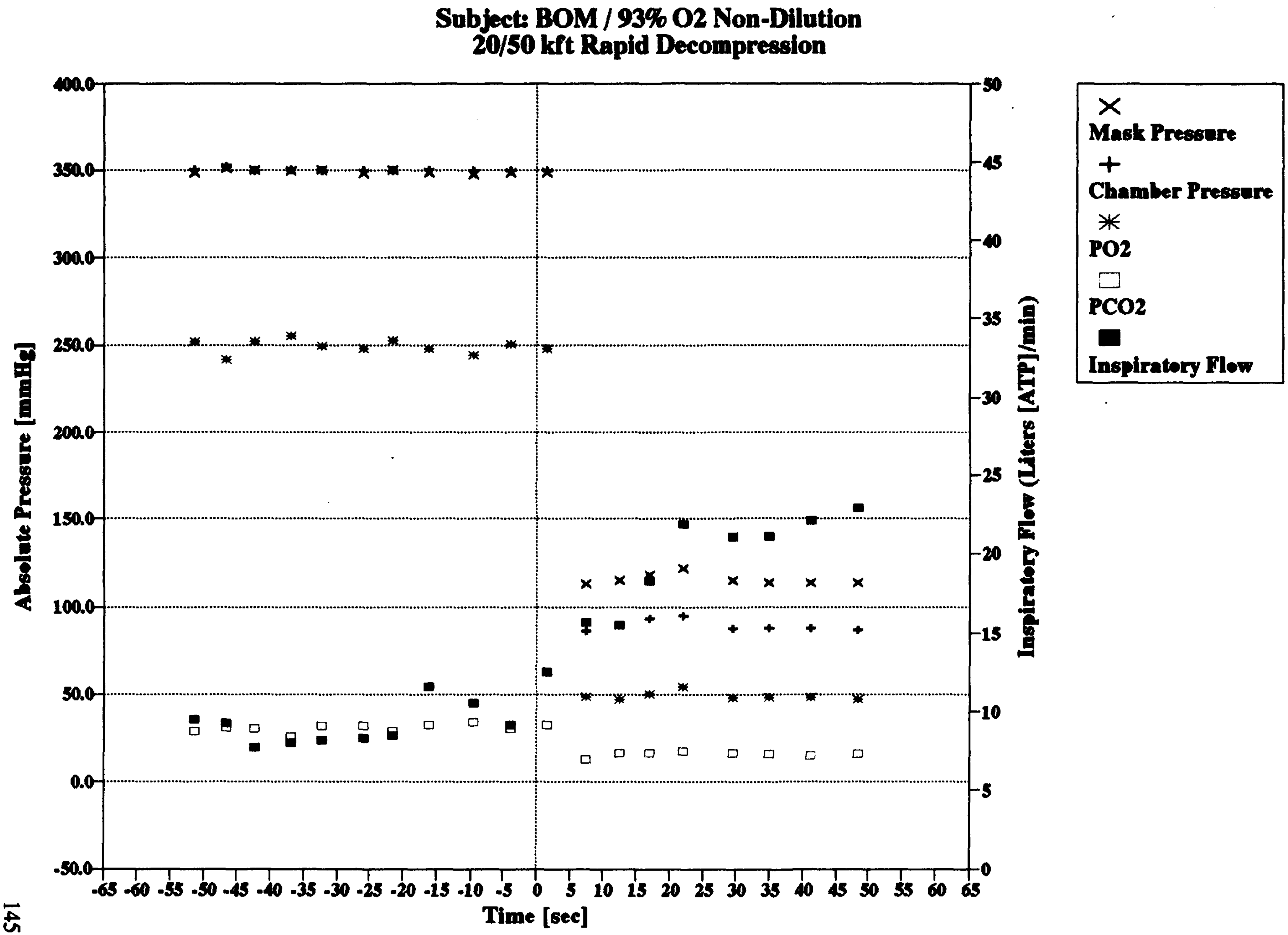

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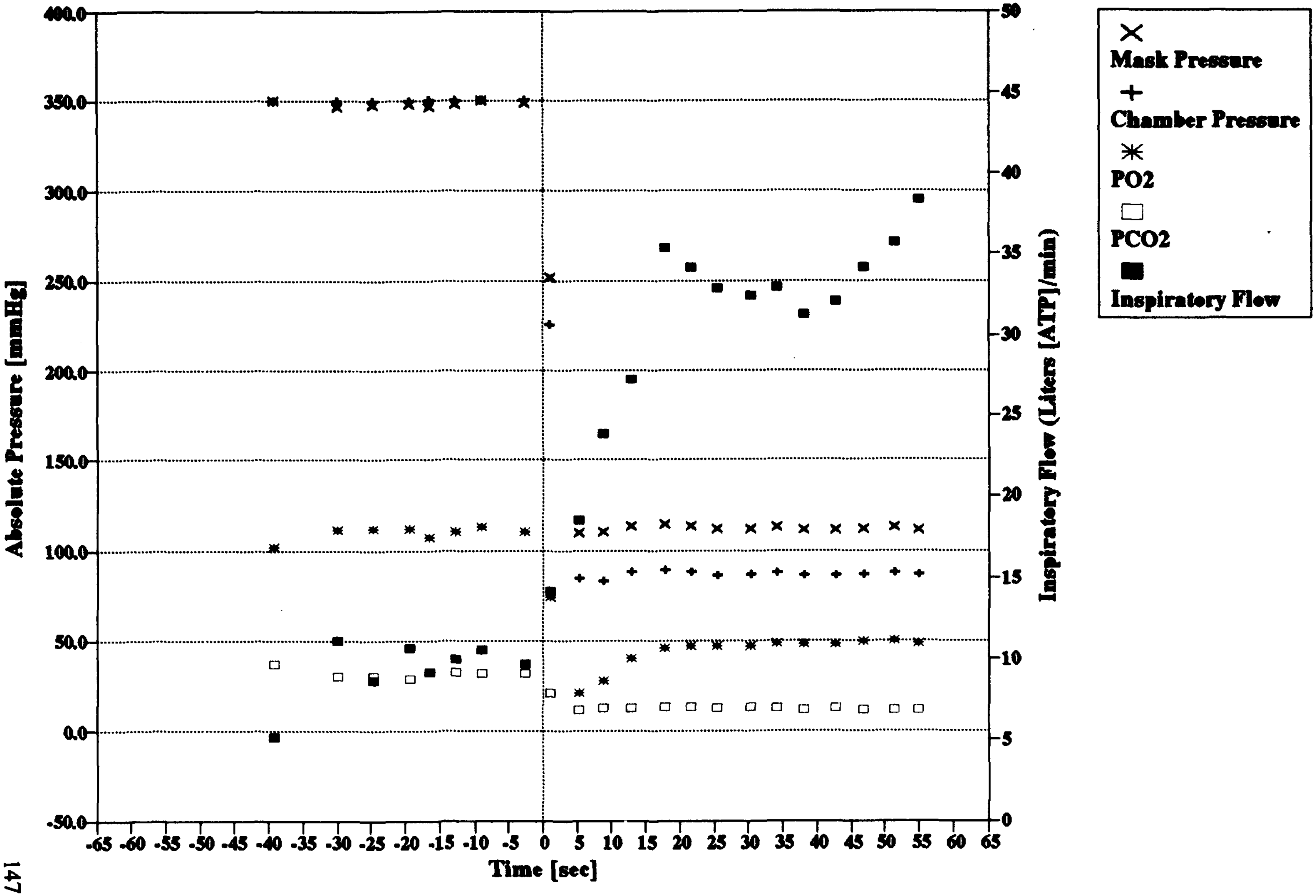
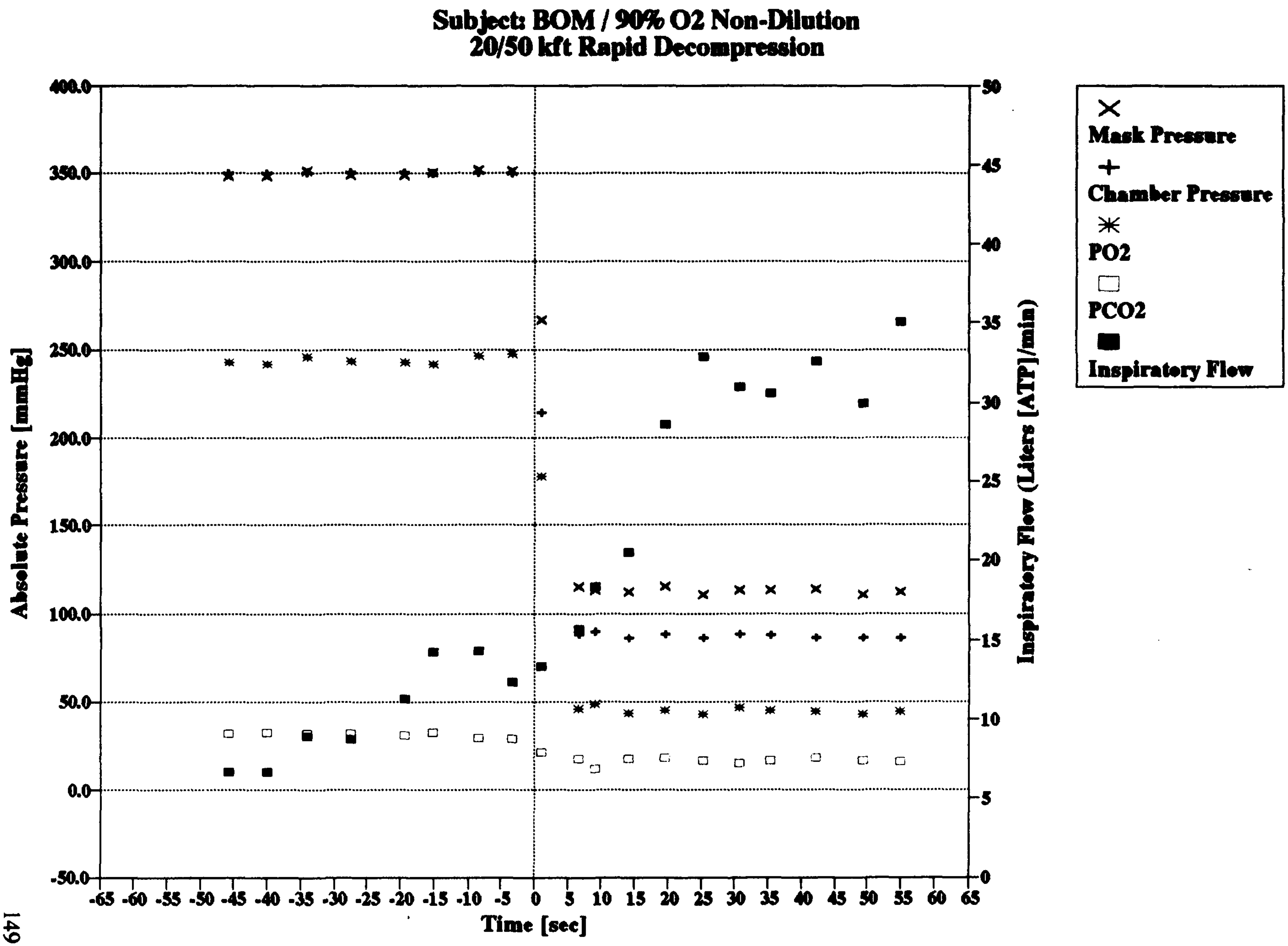
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: BOM / 93% O2 Dilution
20/50 kft Rapid Decompression





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300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
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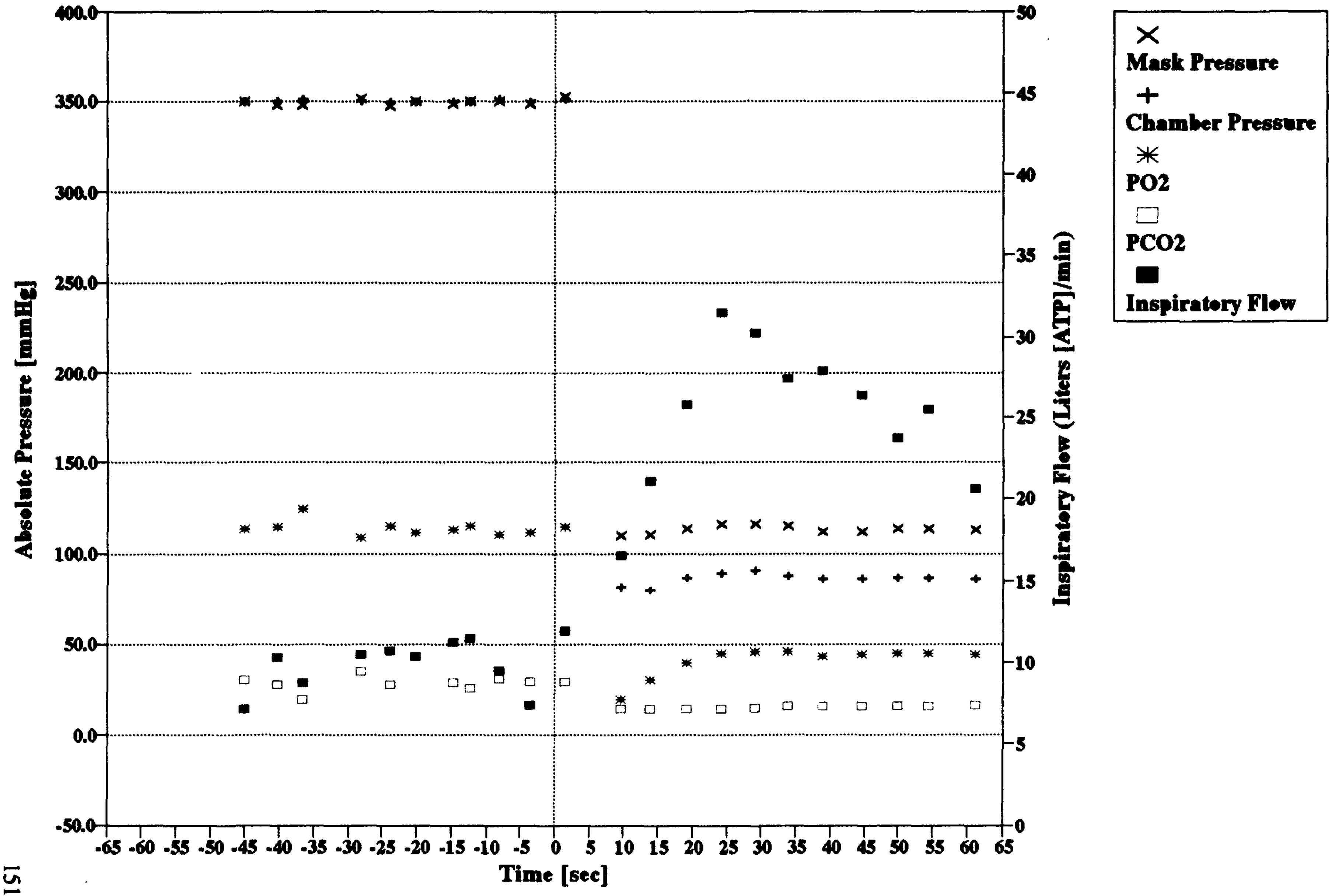
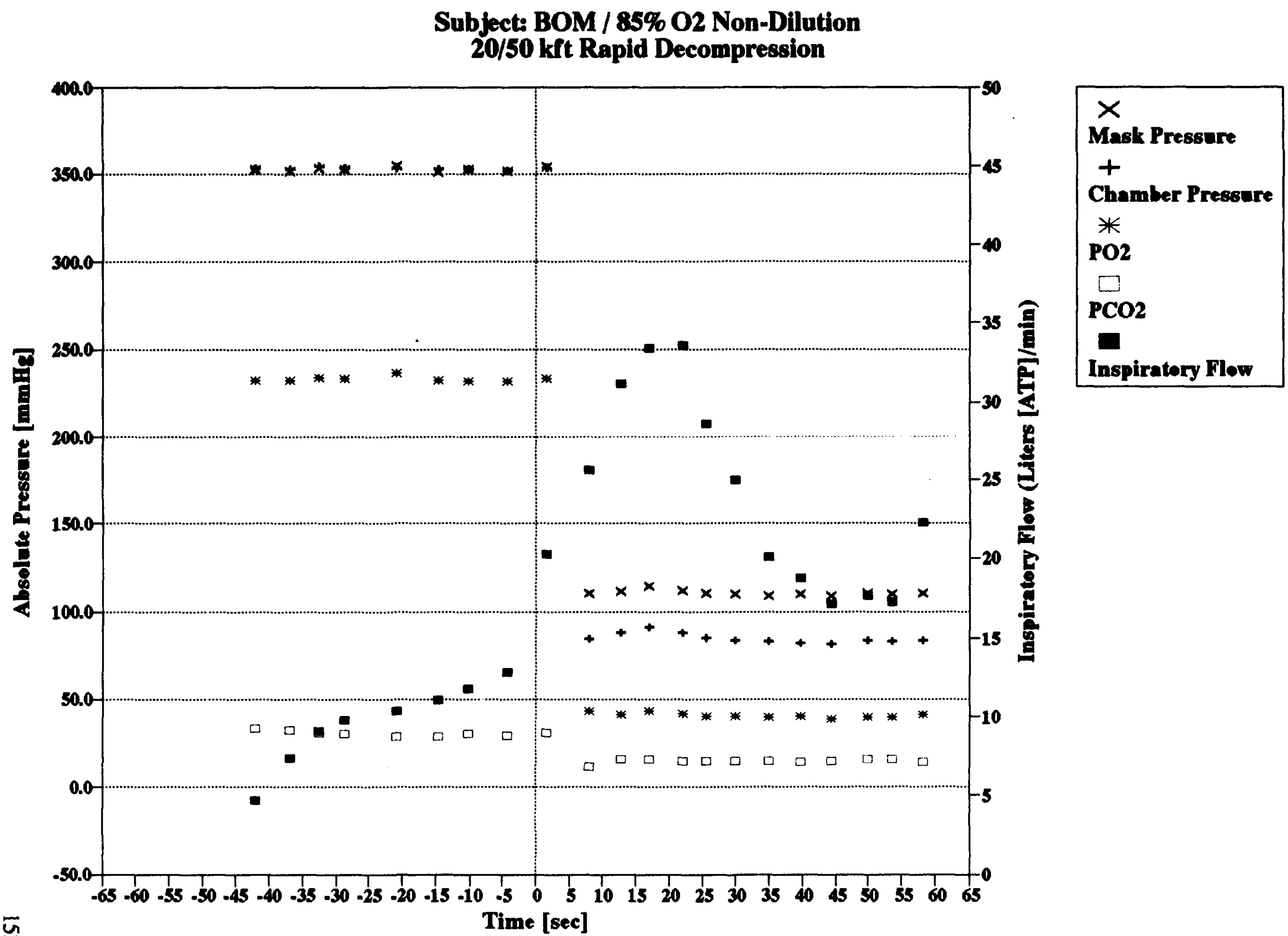
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: BOM / 90% O2 Dilution
20/50 kft Rapid Decompression





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200.0
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100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
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30
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0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

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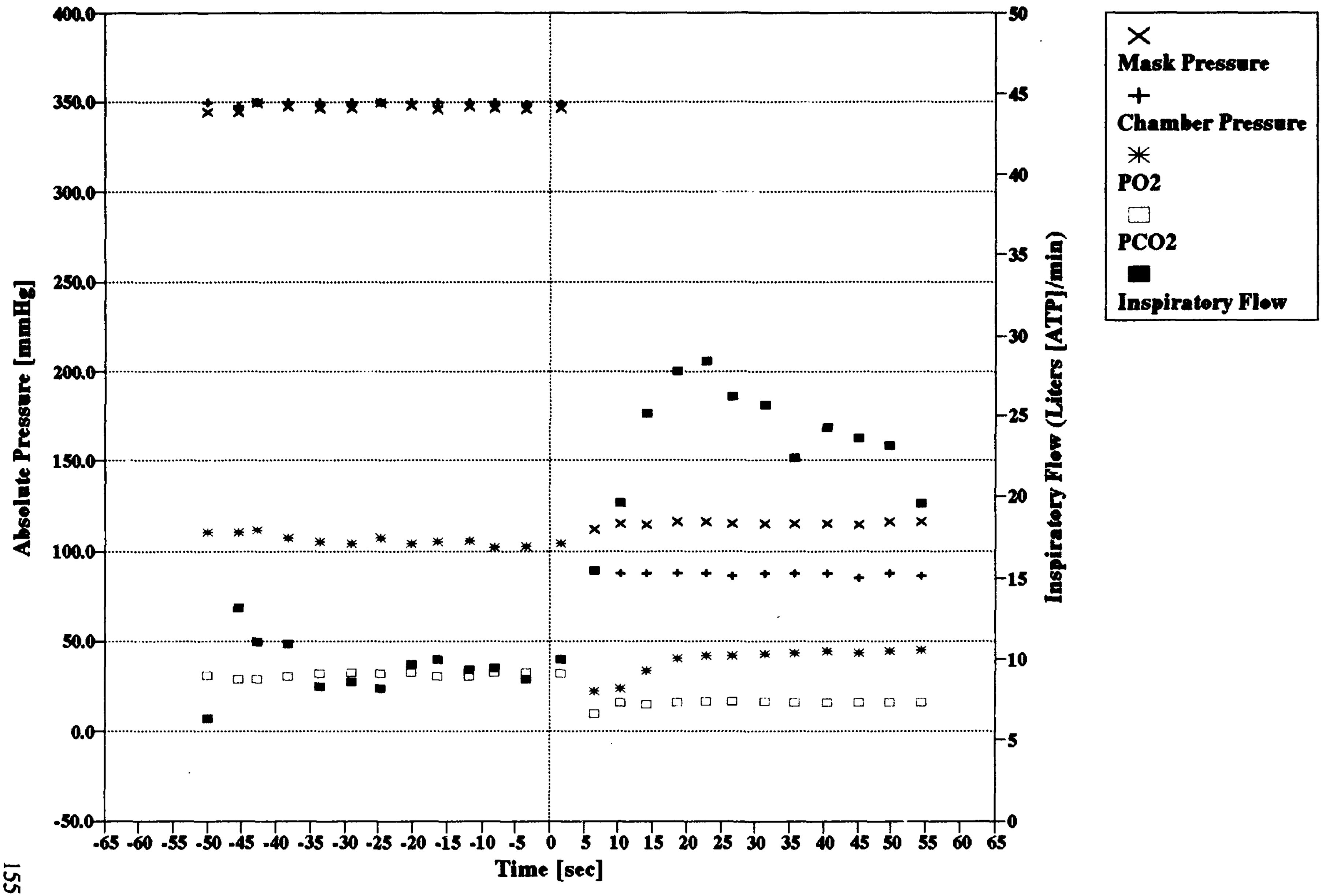
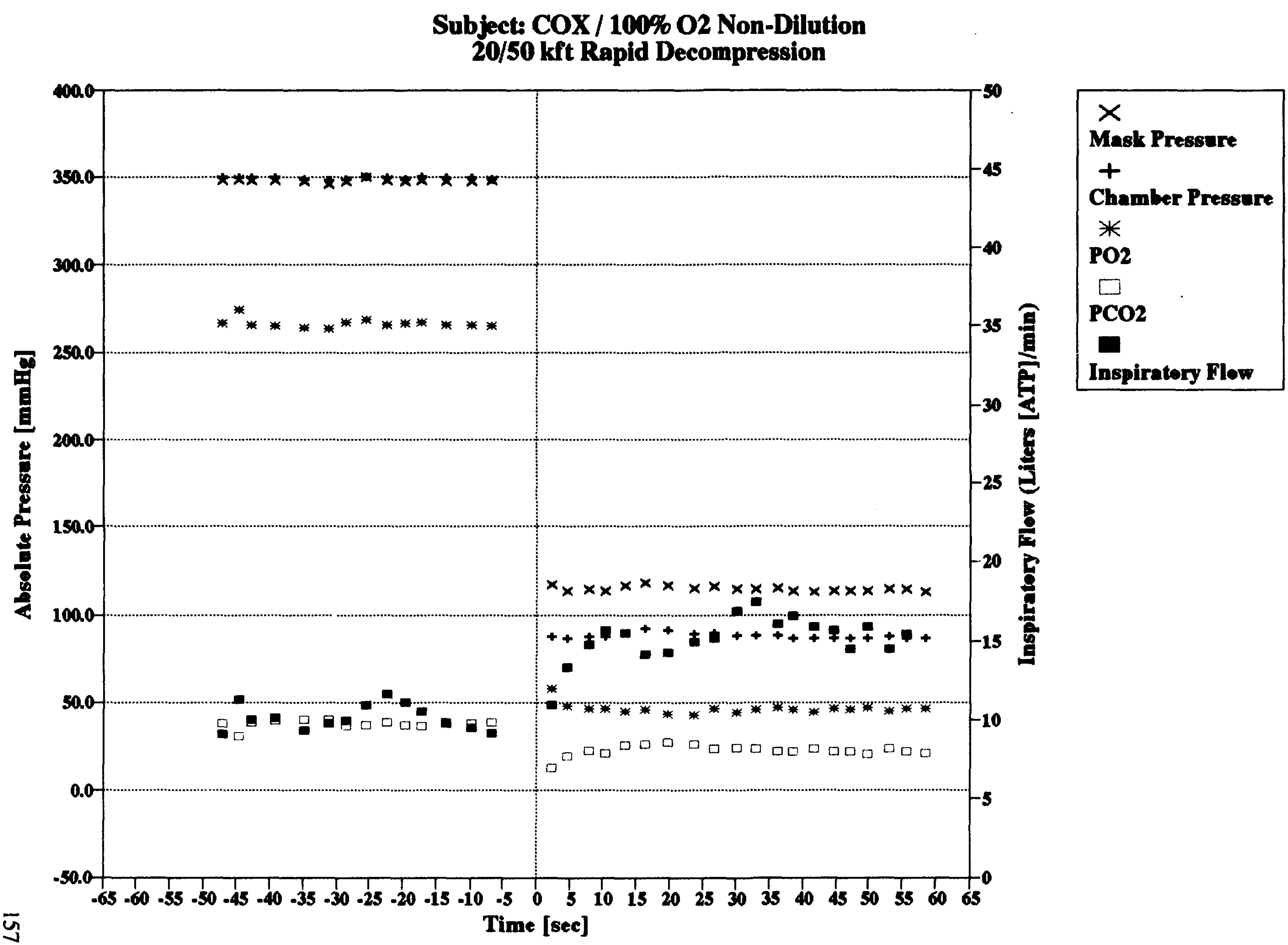
20/50 kft Rapid Decompression





Subject: COX / 100% O2 Non-Dilution
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350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
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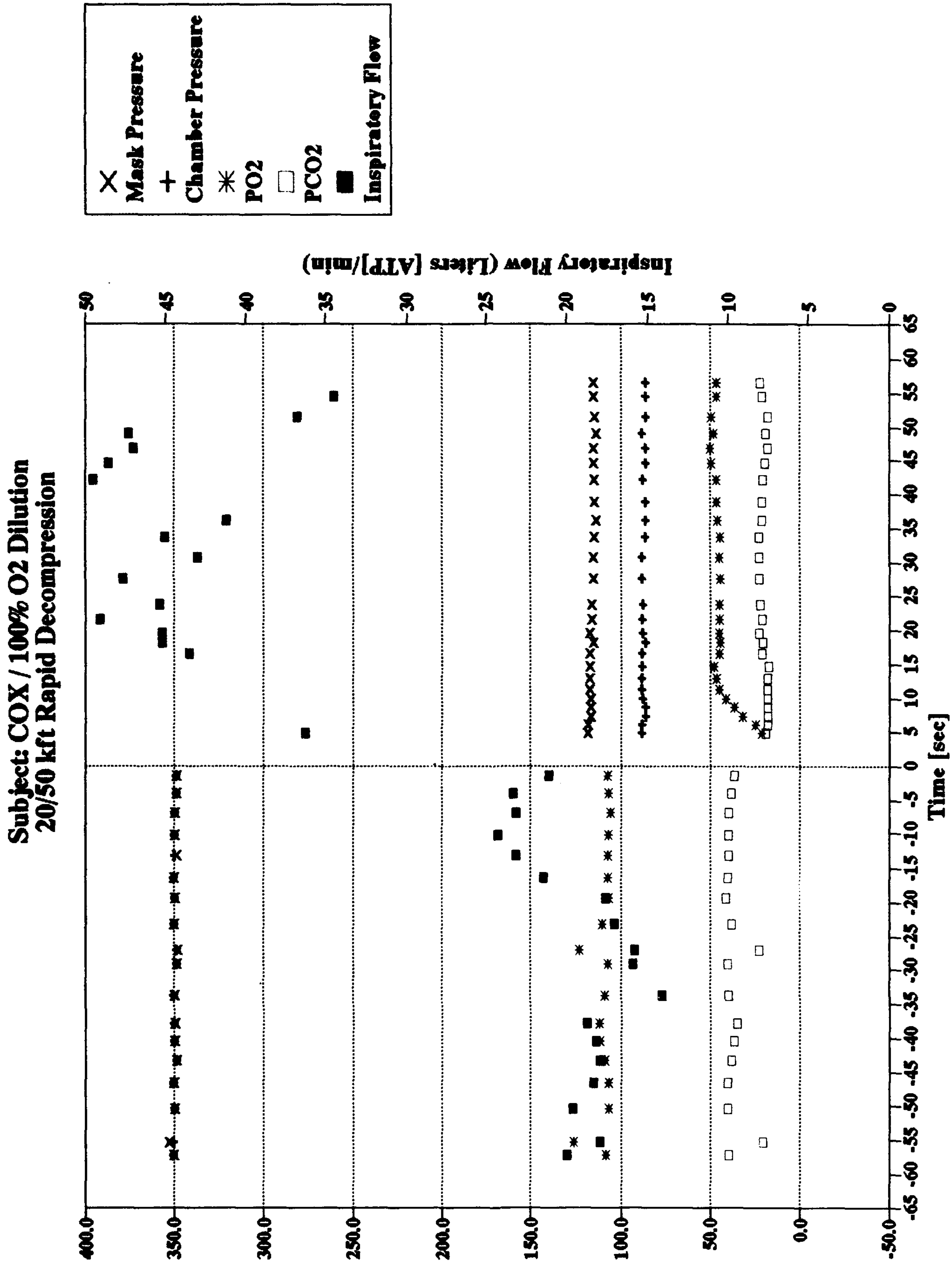
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow



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Chamber Pressure
PO2
PCO2
Inspiratory Flow
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
400.0
350.0
300.0
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200.0
150.0
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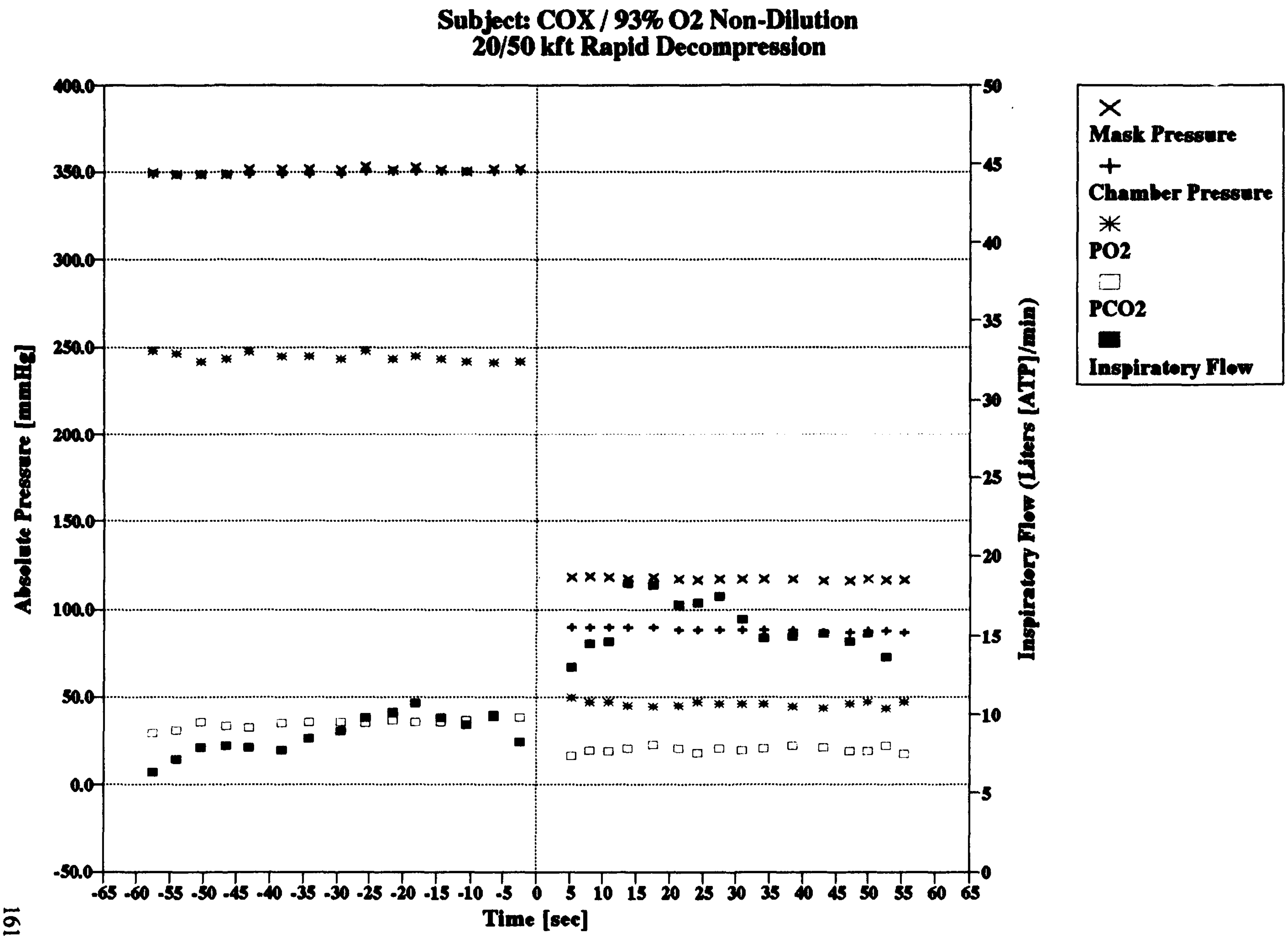

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Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
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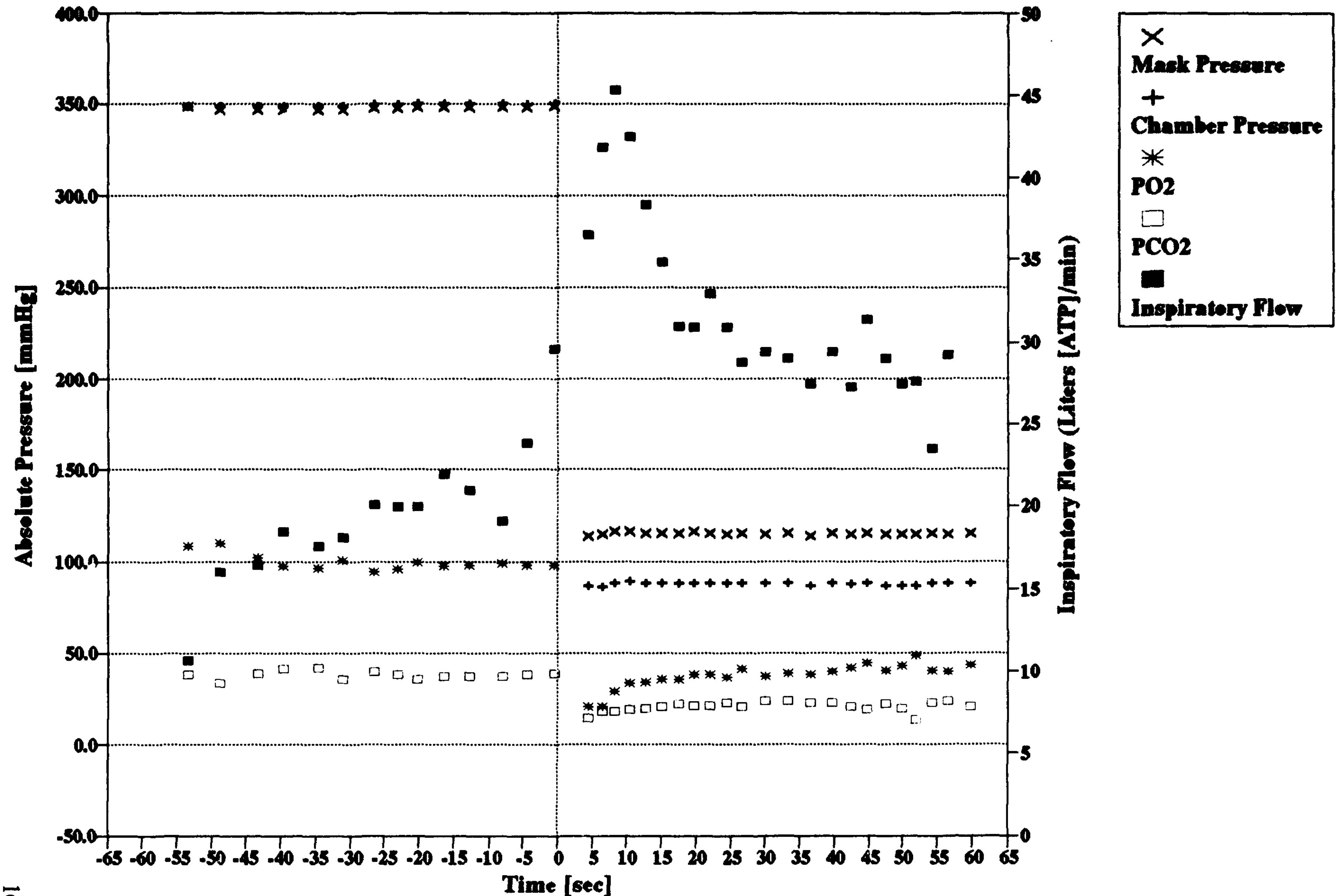
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: COX / 93% O2 Dilution
20/50 kft Rapid Decompression



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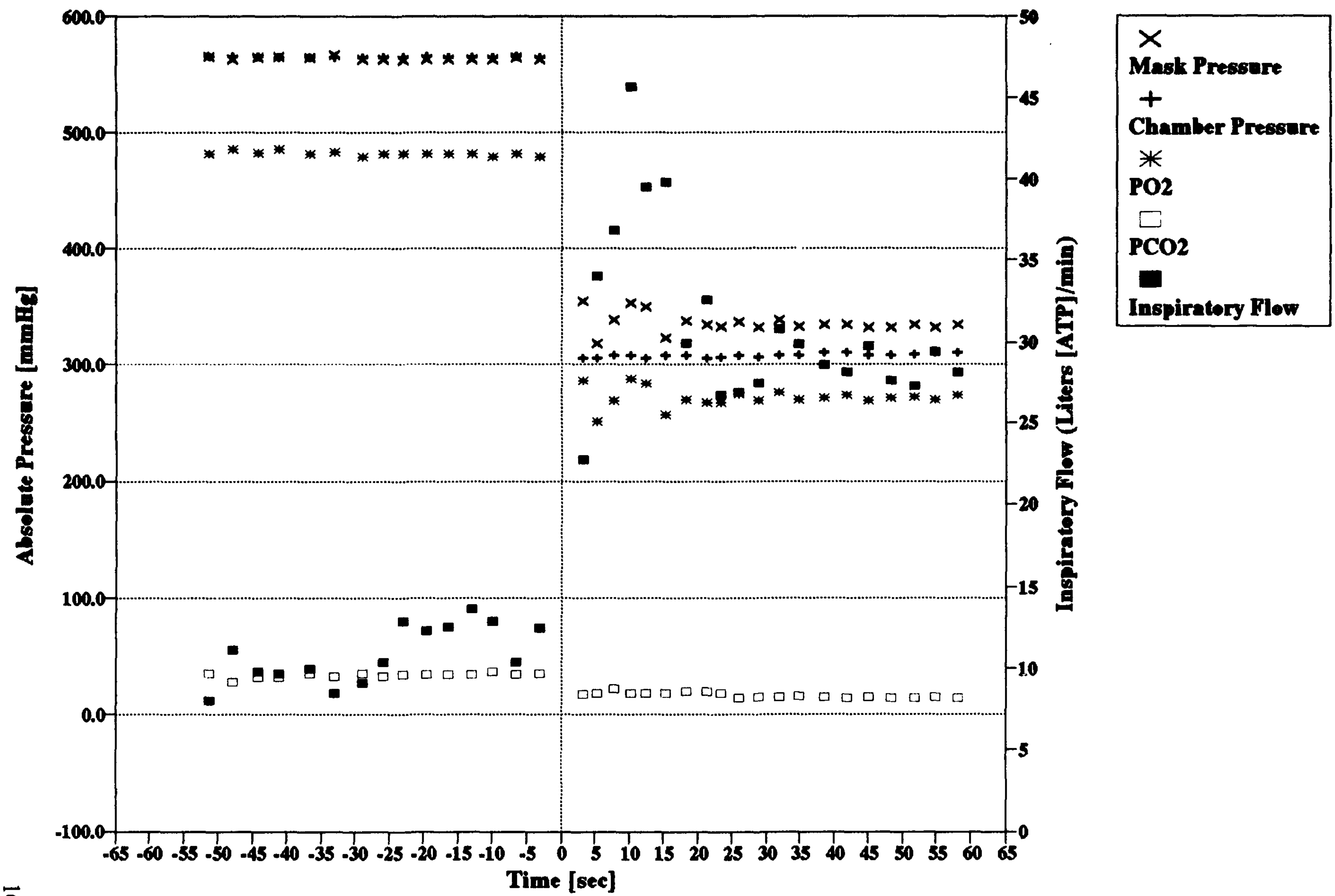

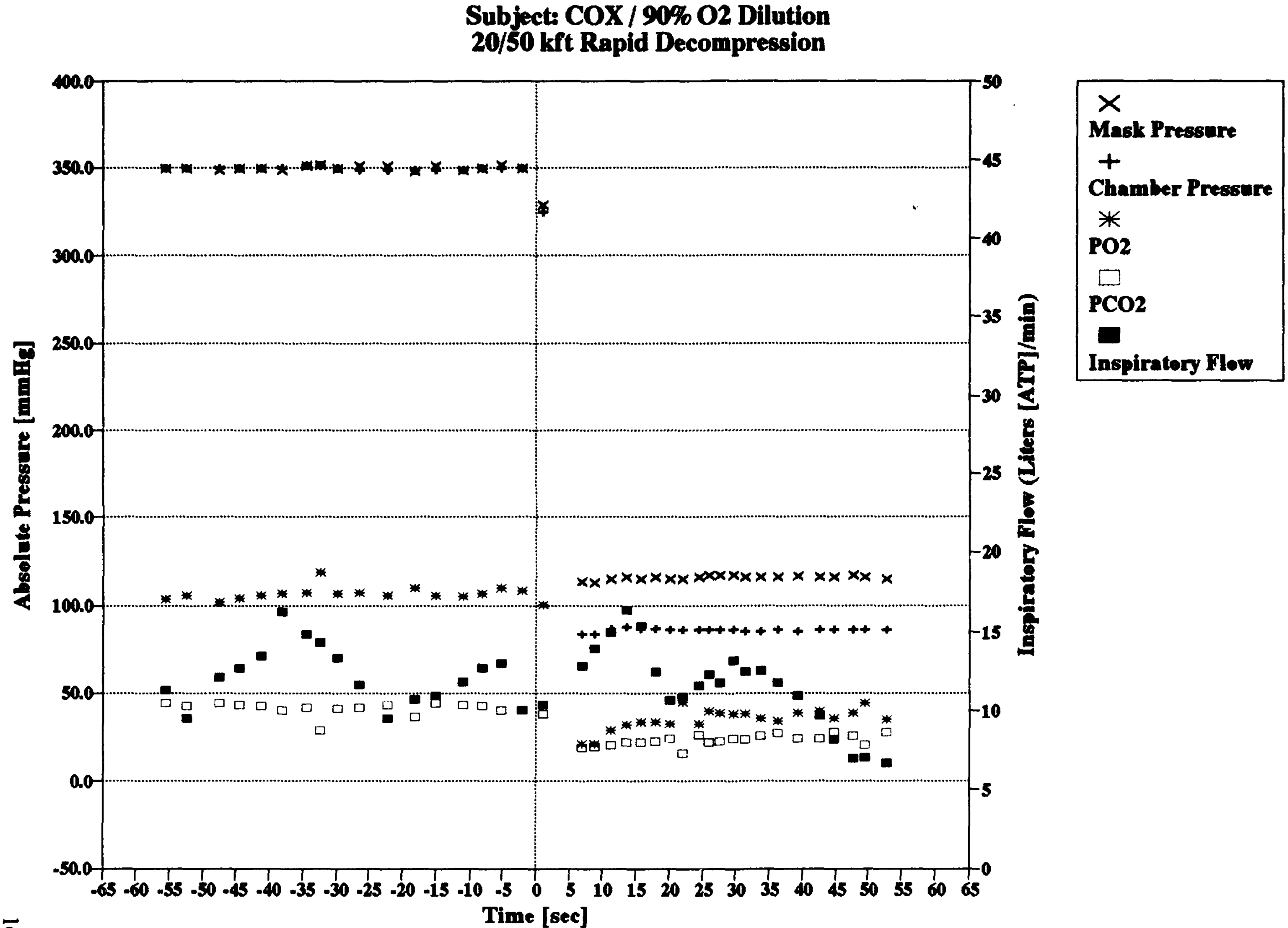

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300.0
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50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
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Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

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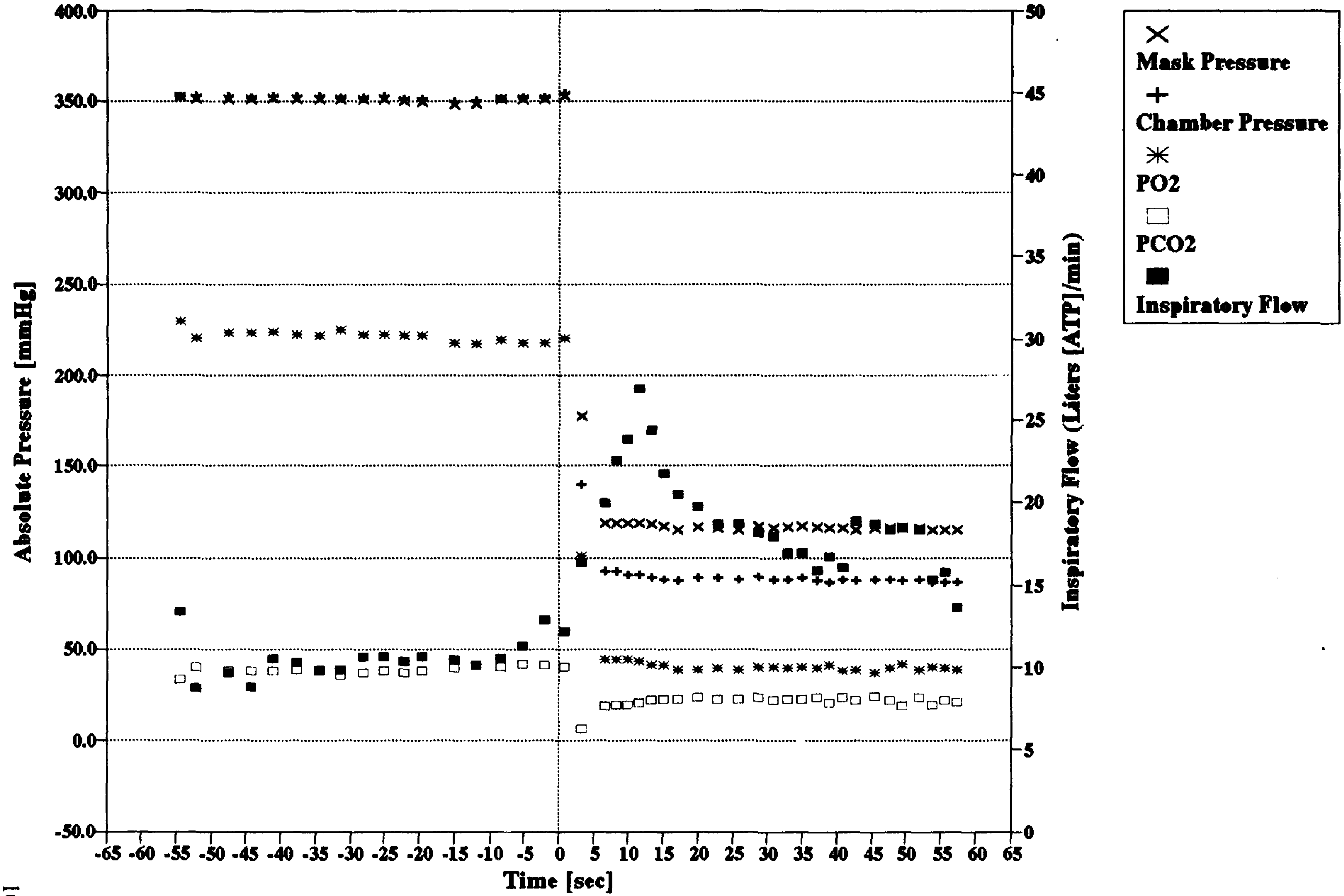

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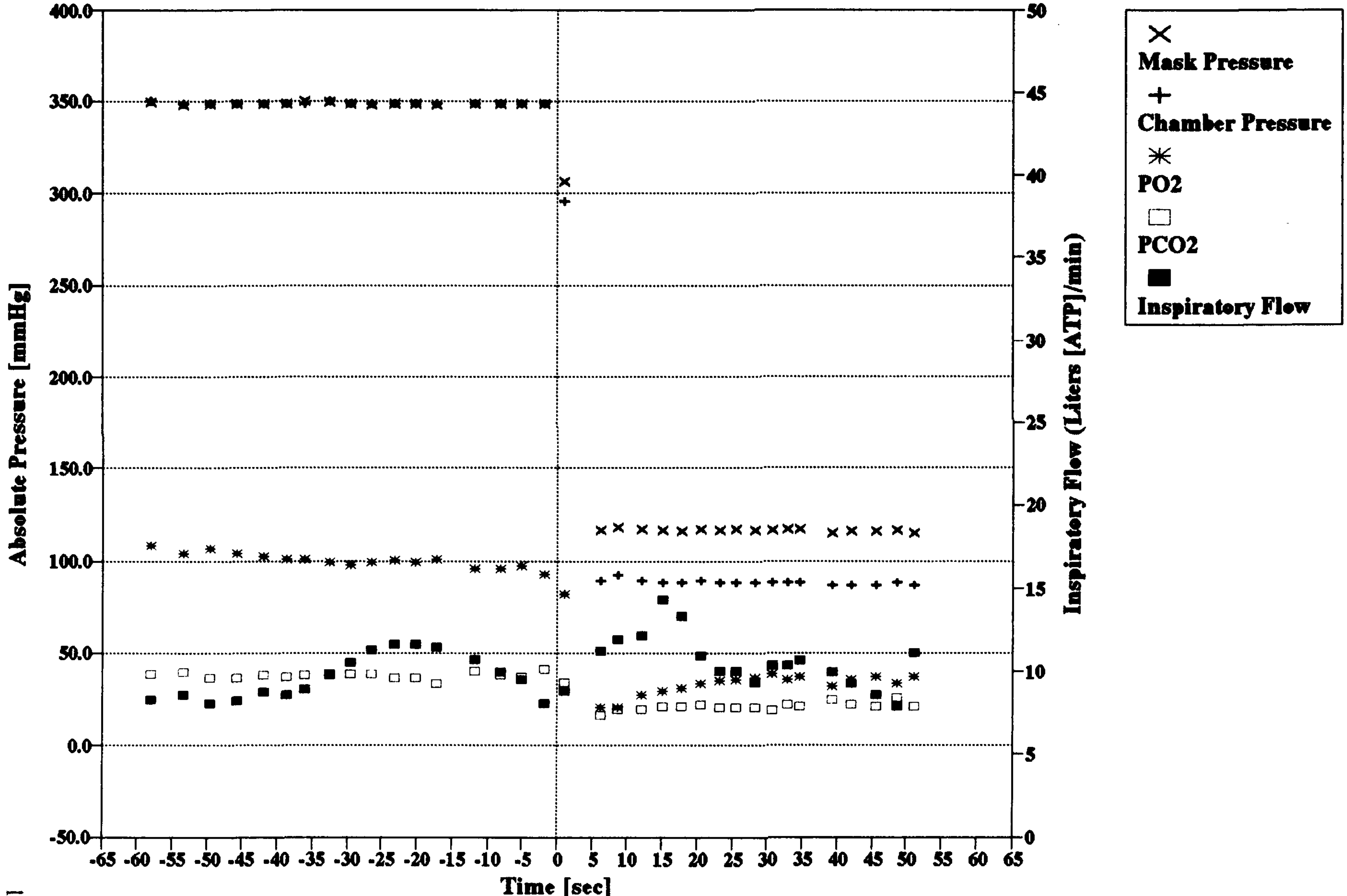

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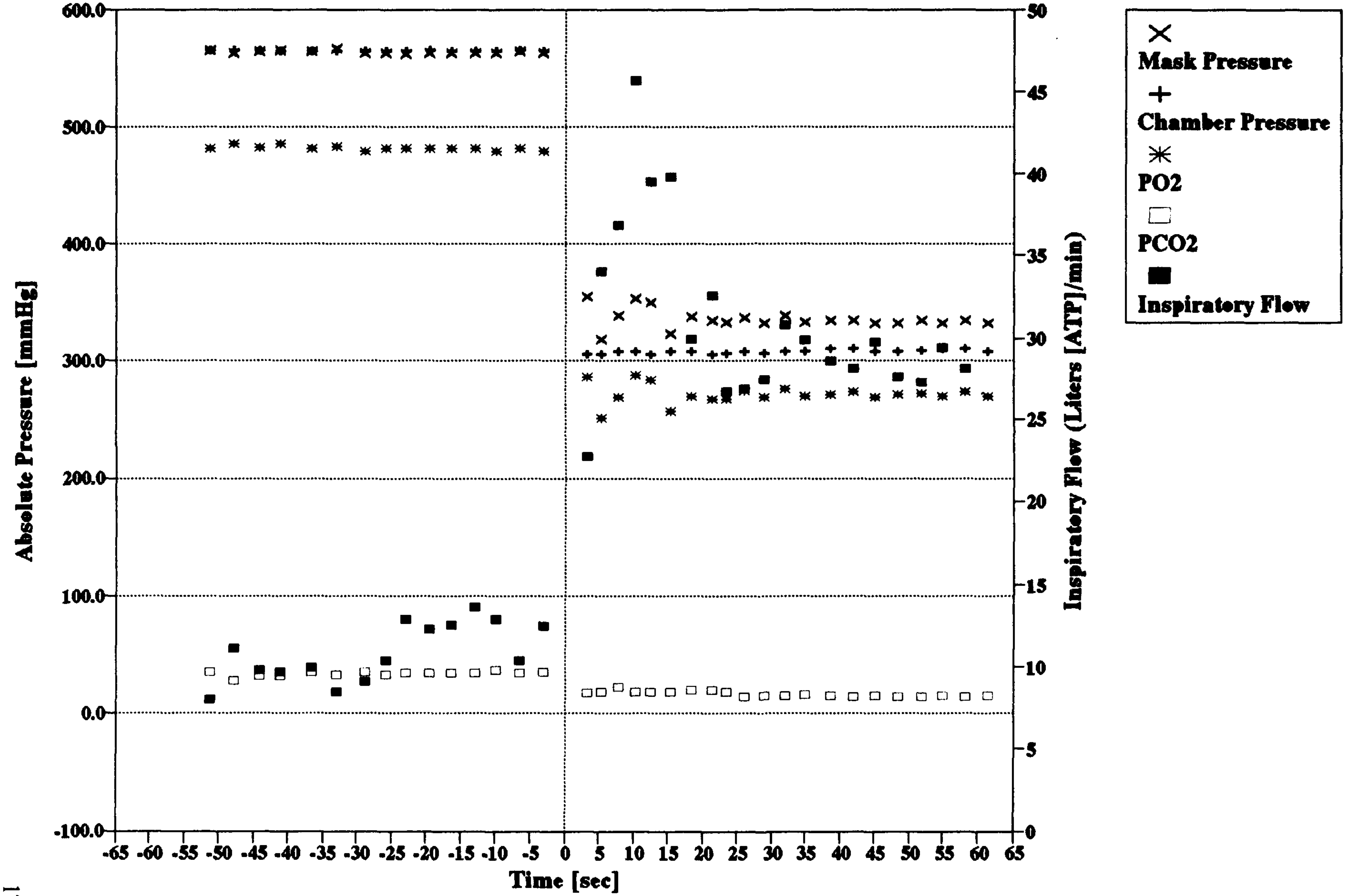

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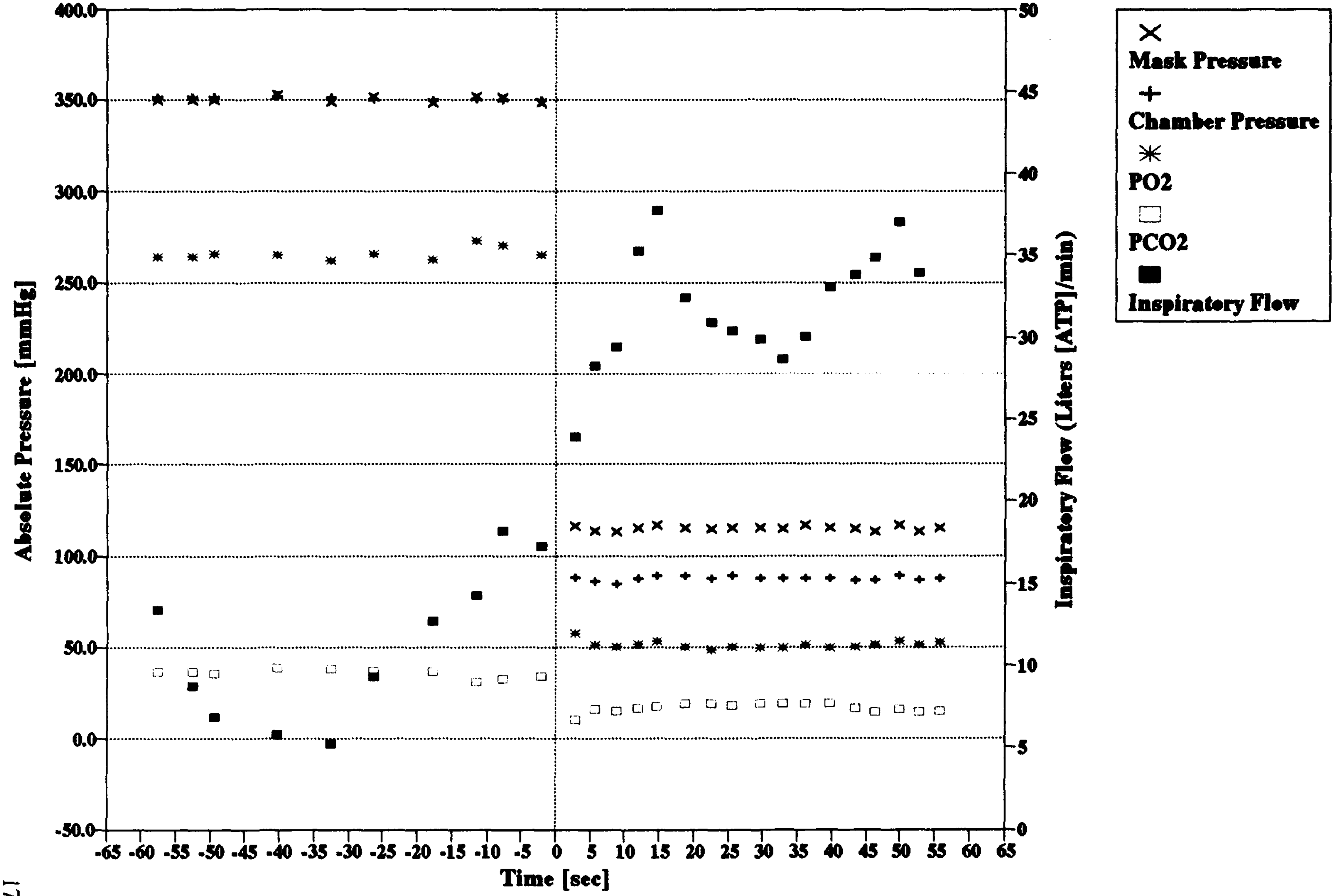

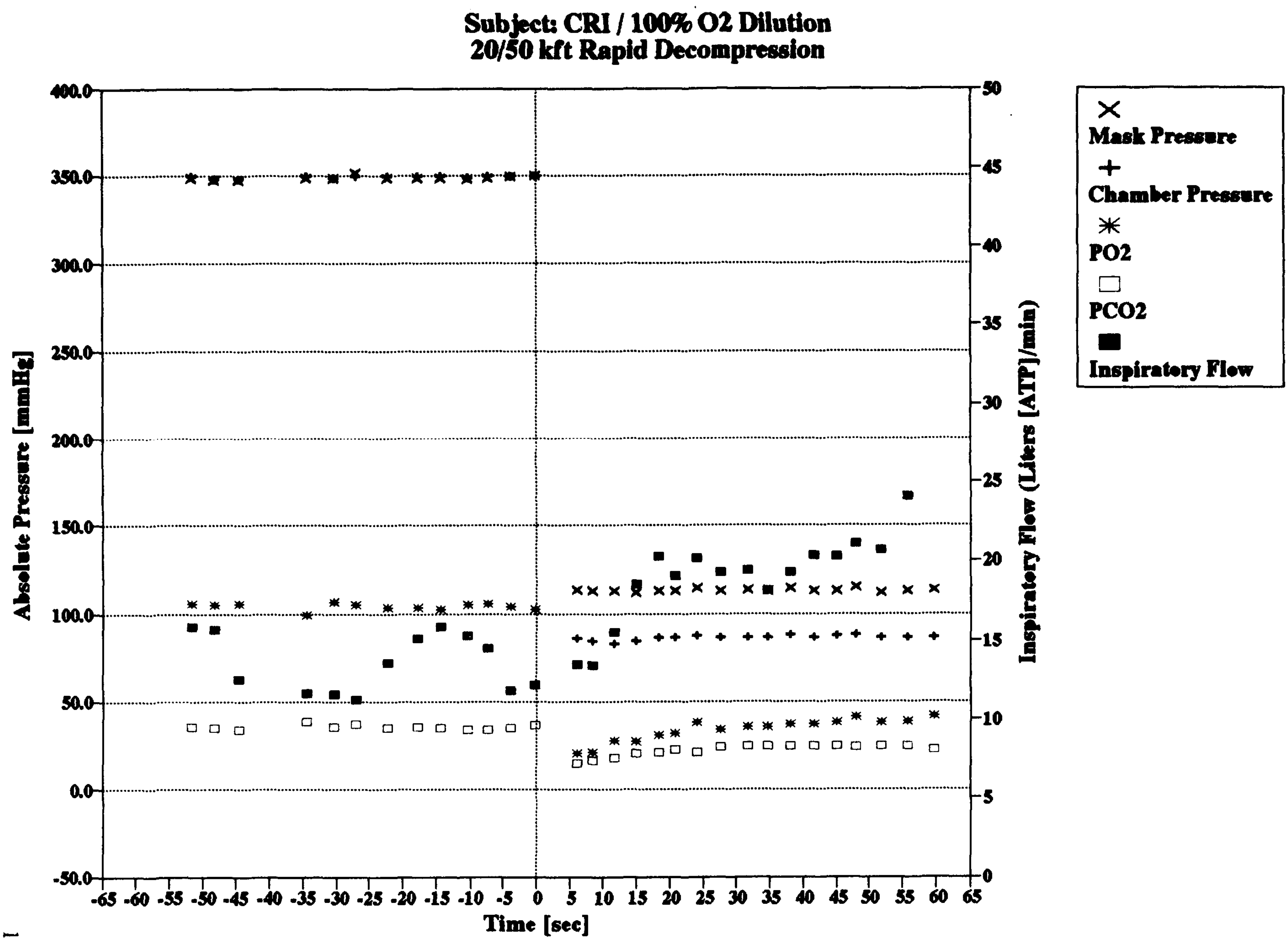
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0.0
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Inspiratory Flow (Liters [ATP]/min)
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-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
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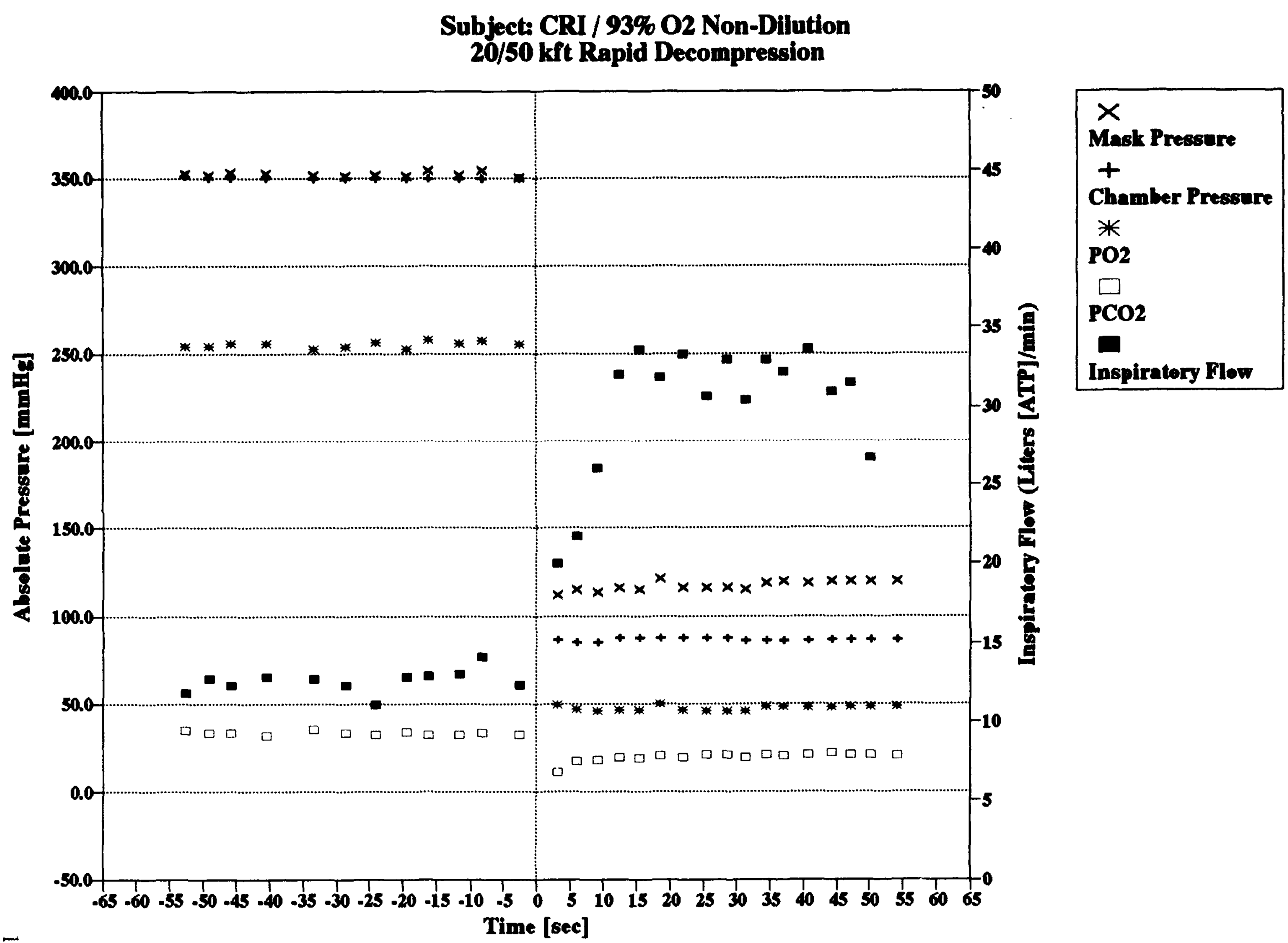
PO2
PCO2
Inspiratory Flow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0.0
-50.0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
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0
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

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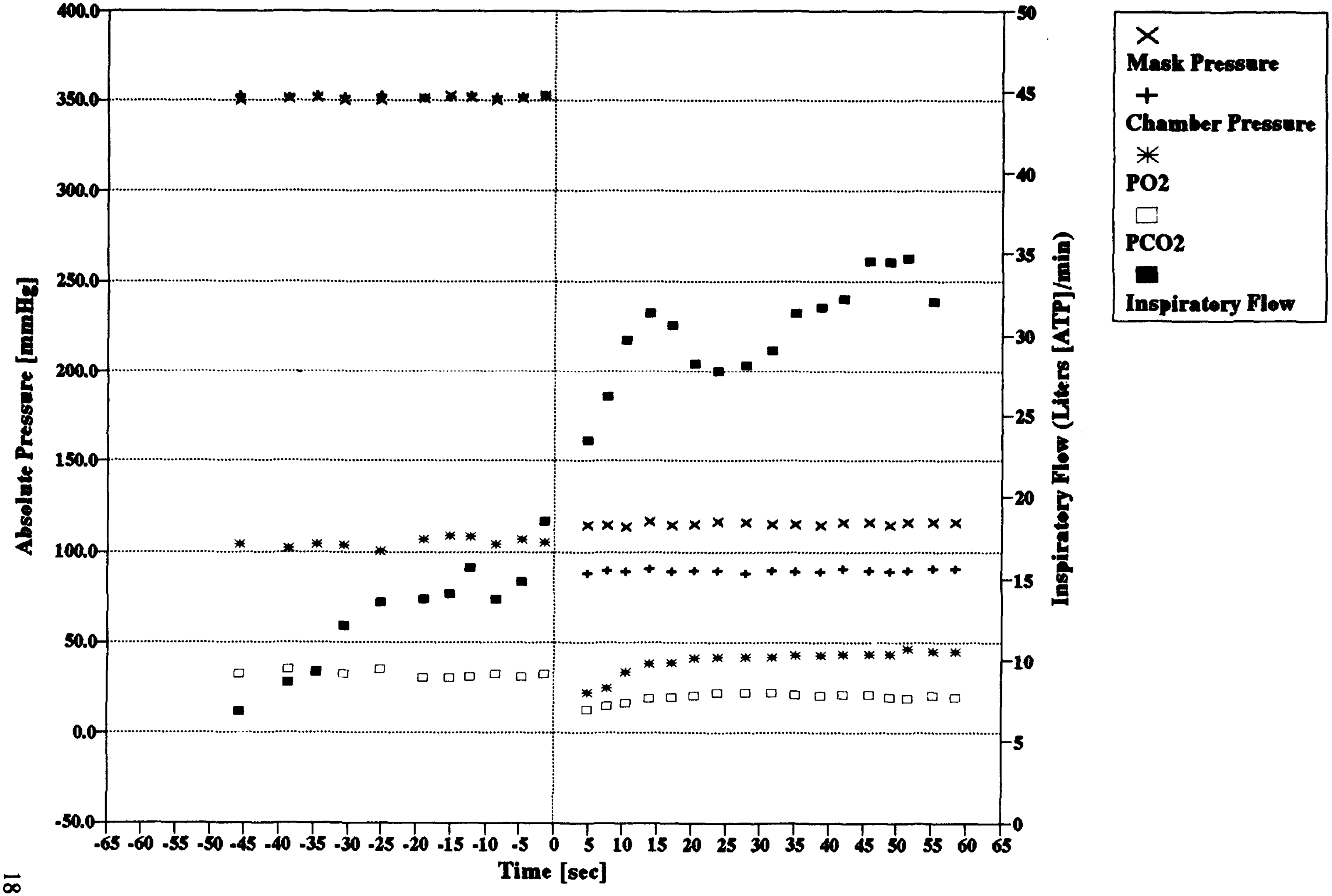
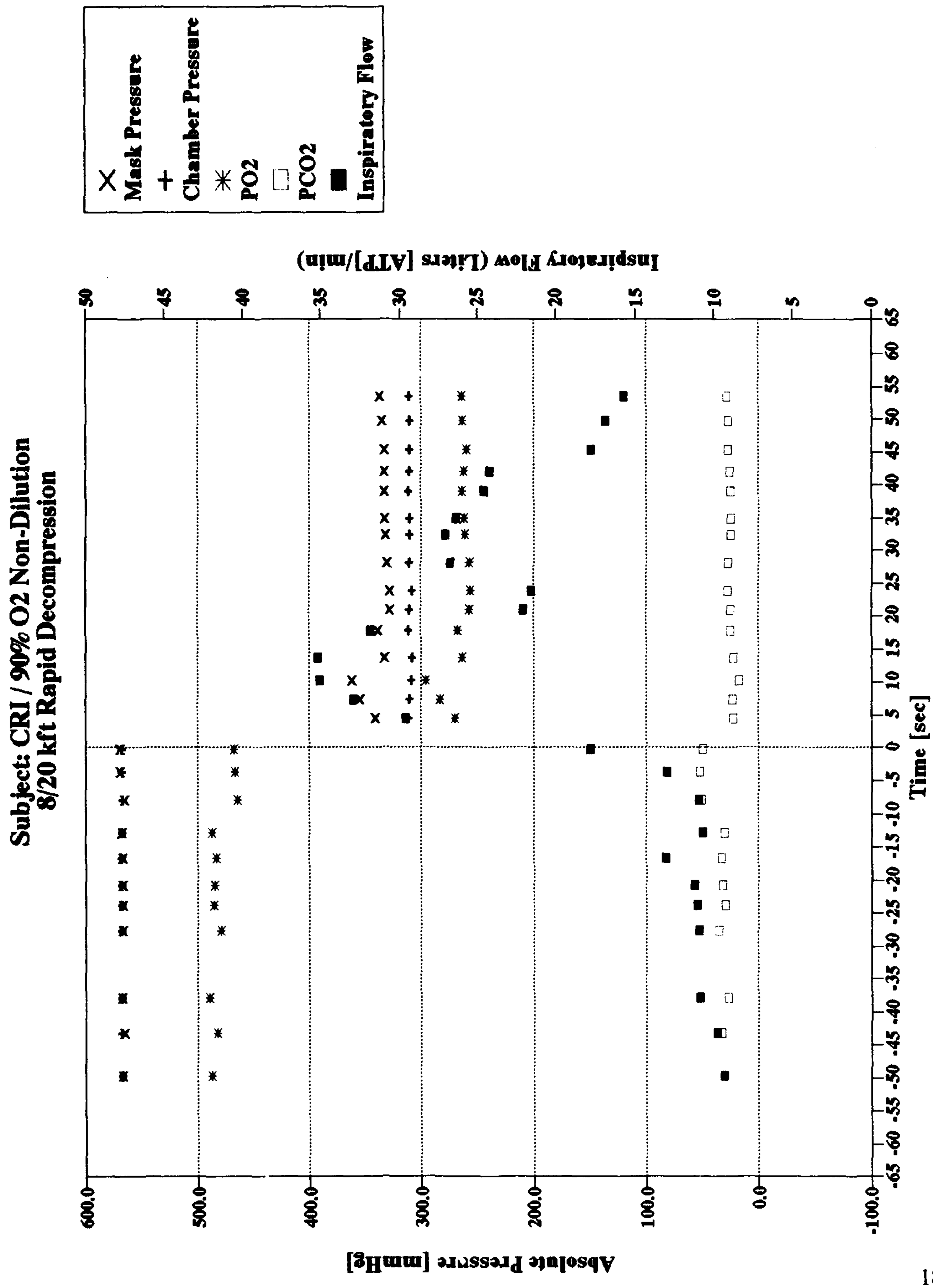
20/50 kft Rapid Decompression





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8/20 kft Rapid Decompression
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow
Absolute Pressure [mmHg]
Time [sec]
Inspiratory Flow (Liters [ATP]/min)

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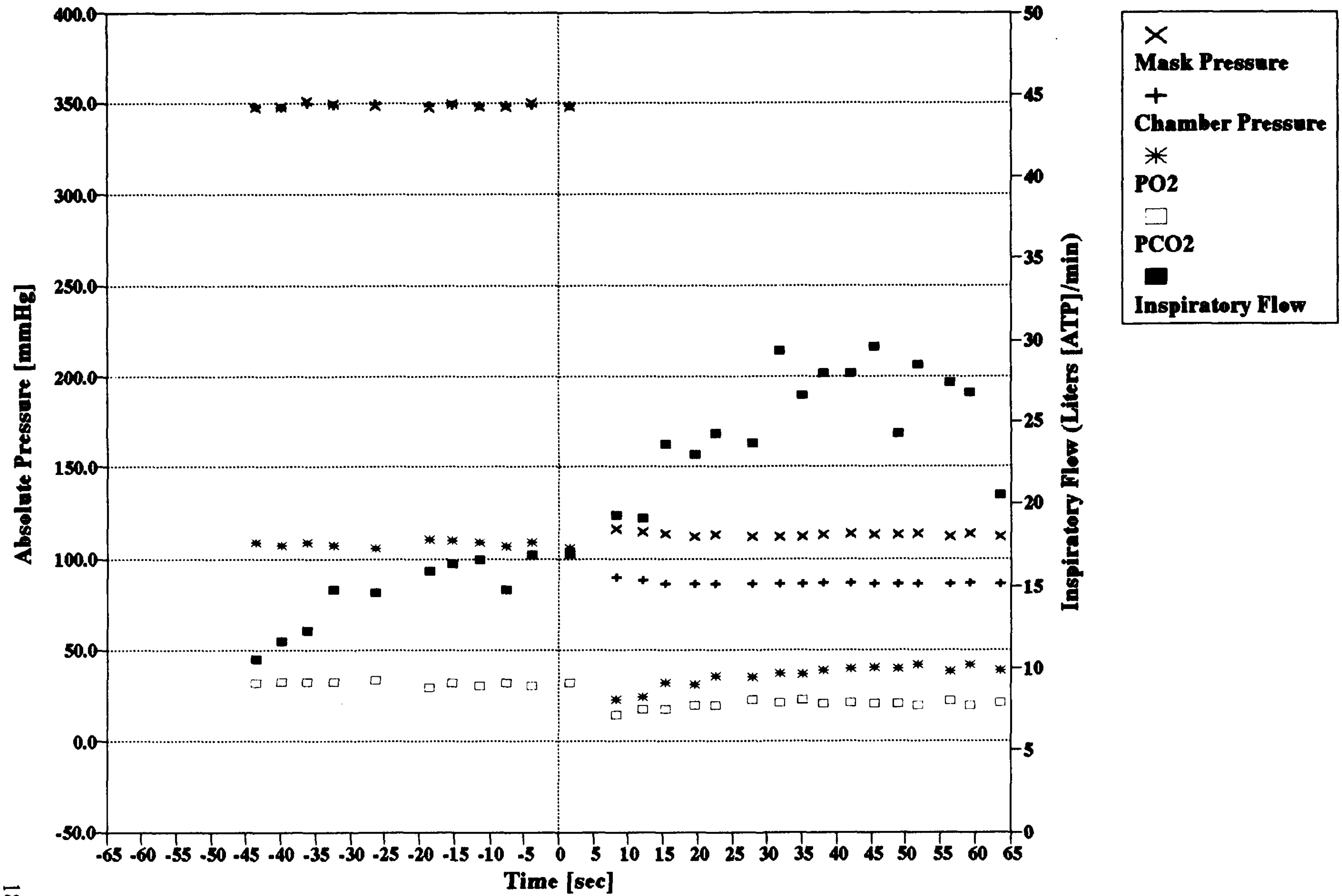
20/50 kft Rapid Decompression



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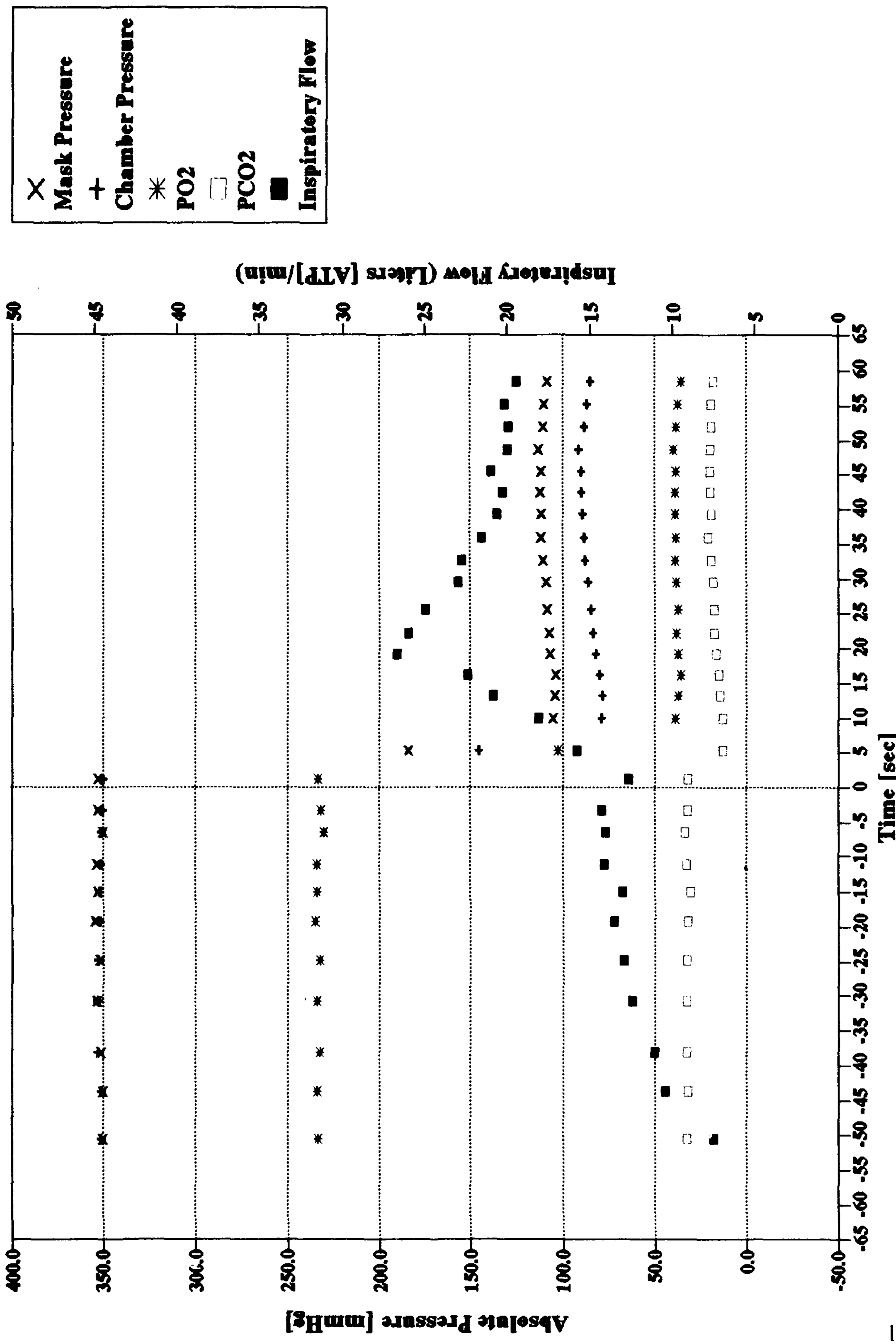

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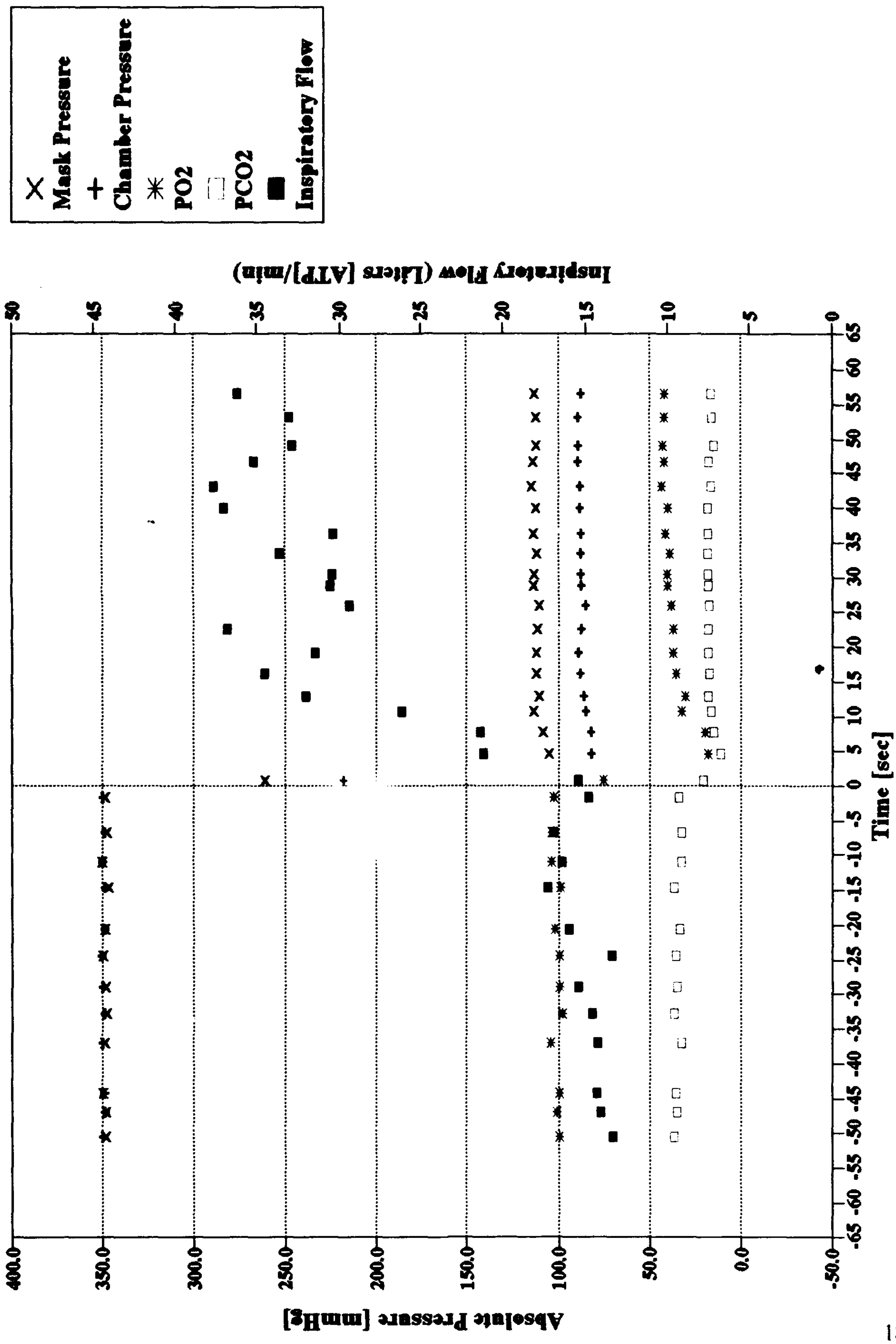

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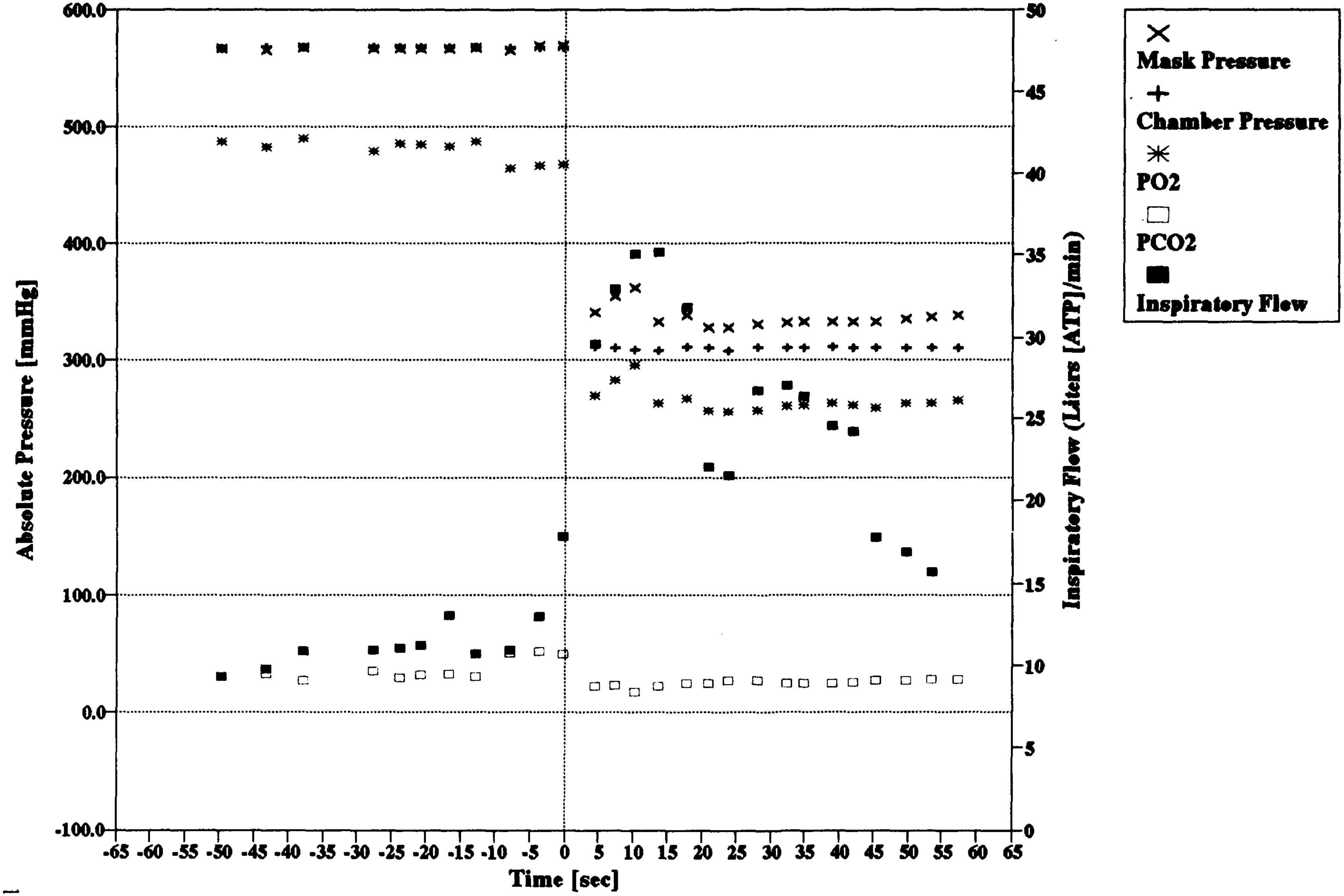

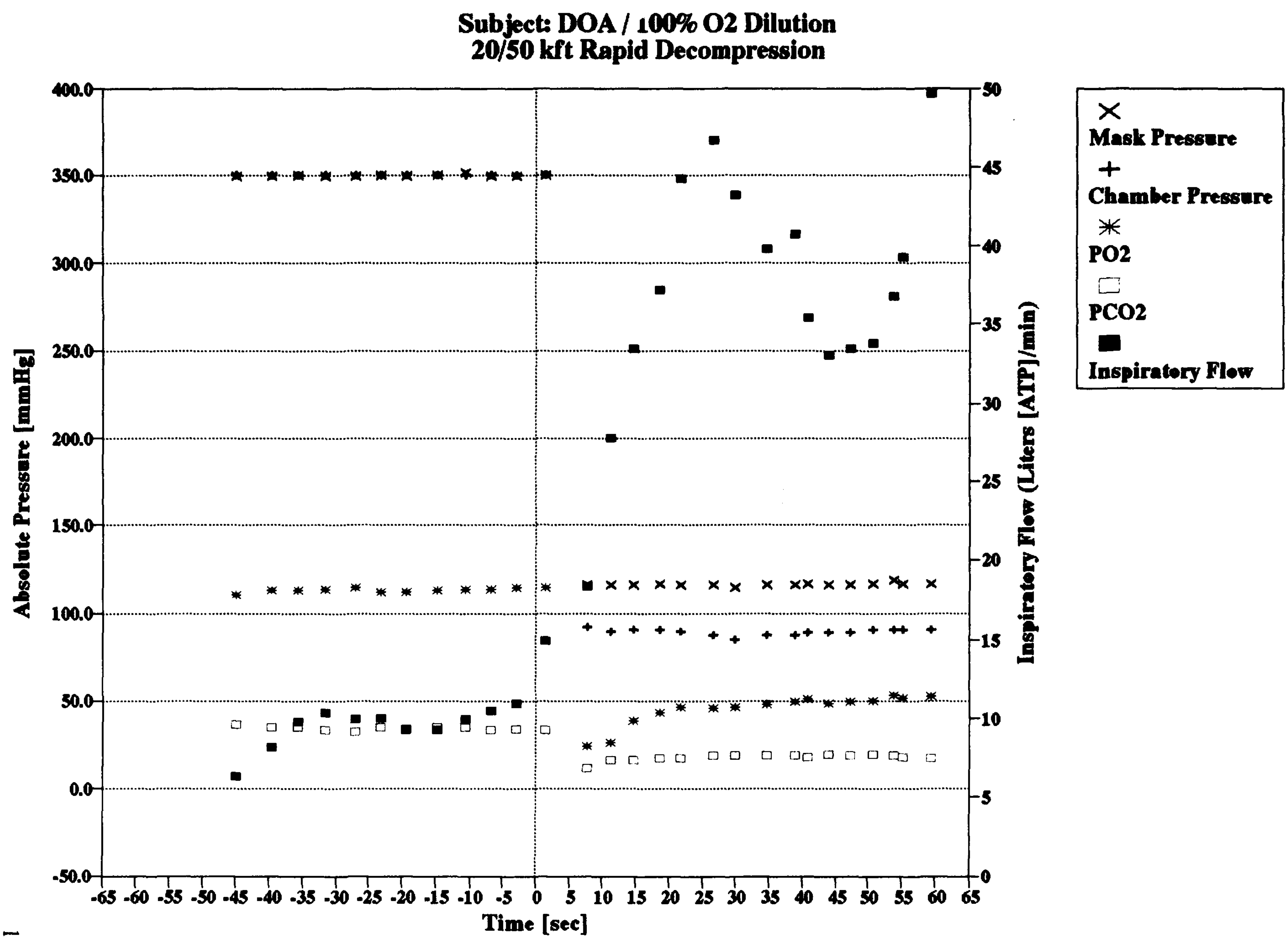

Subject: DOA / 100% O2 Dilution
20/50 kft Rapid Decompression
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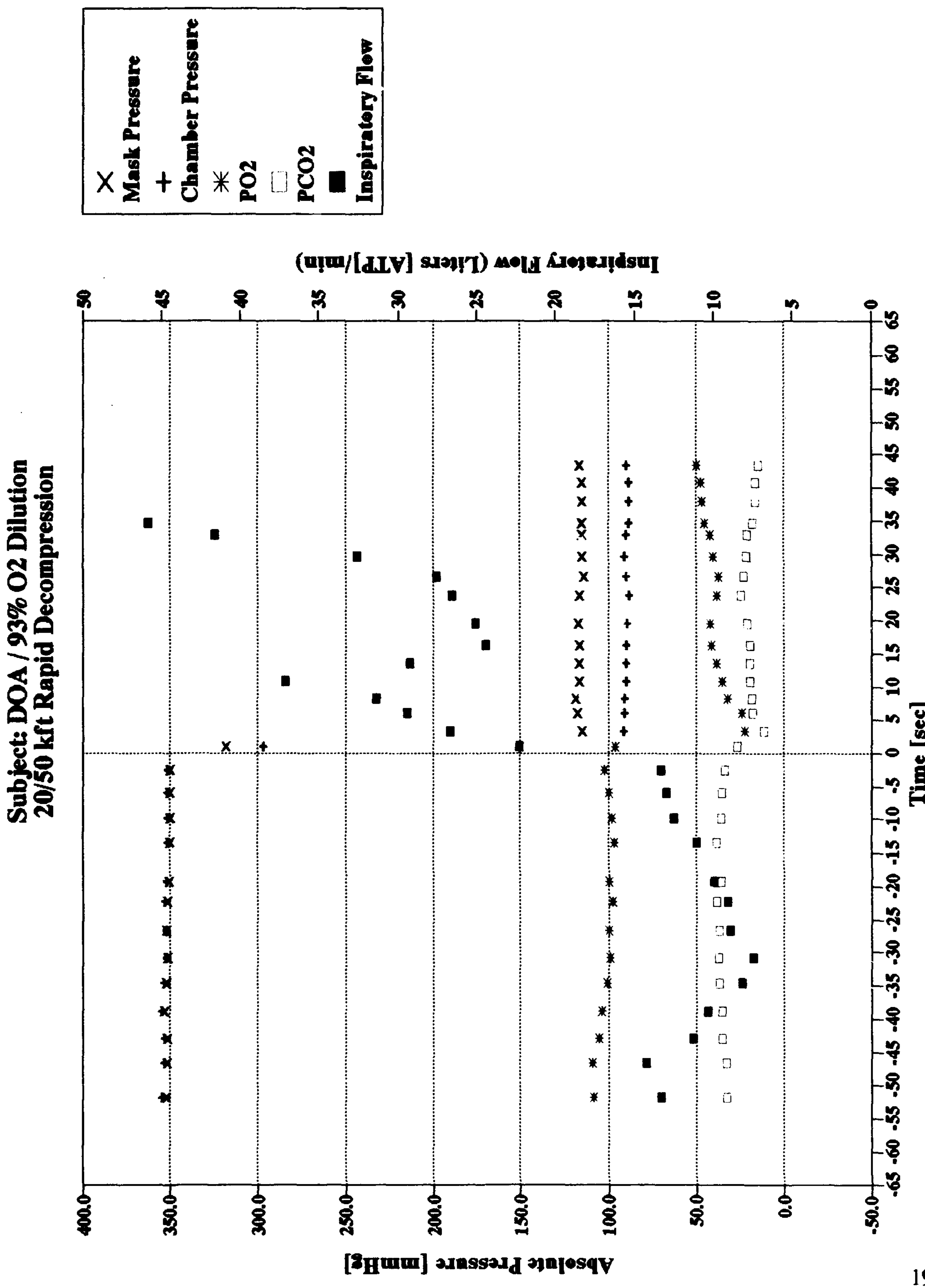
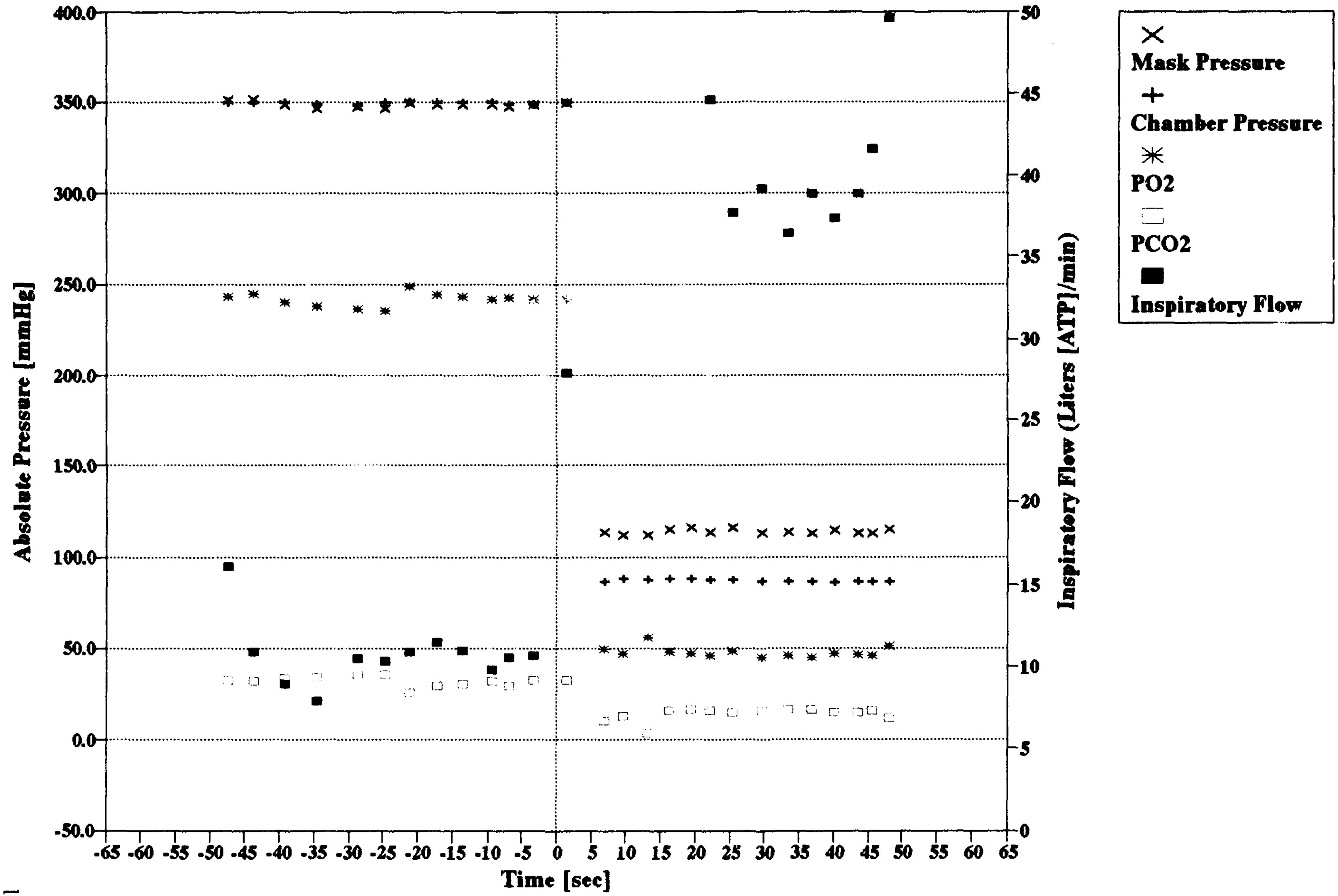
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

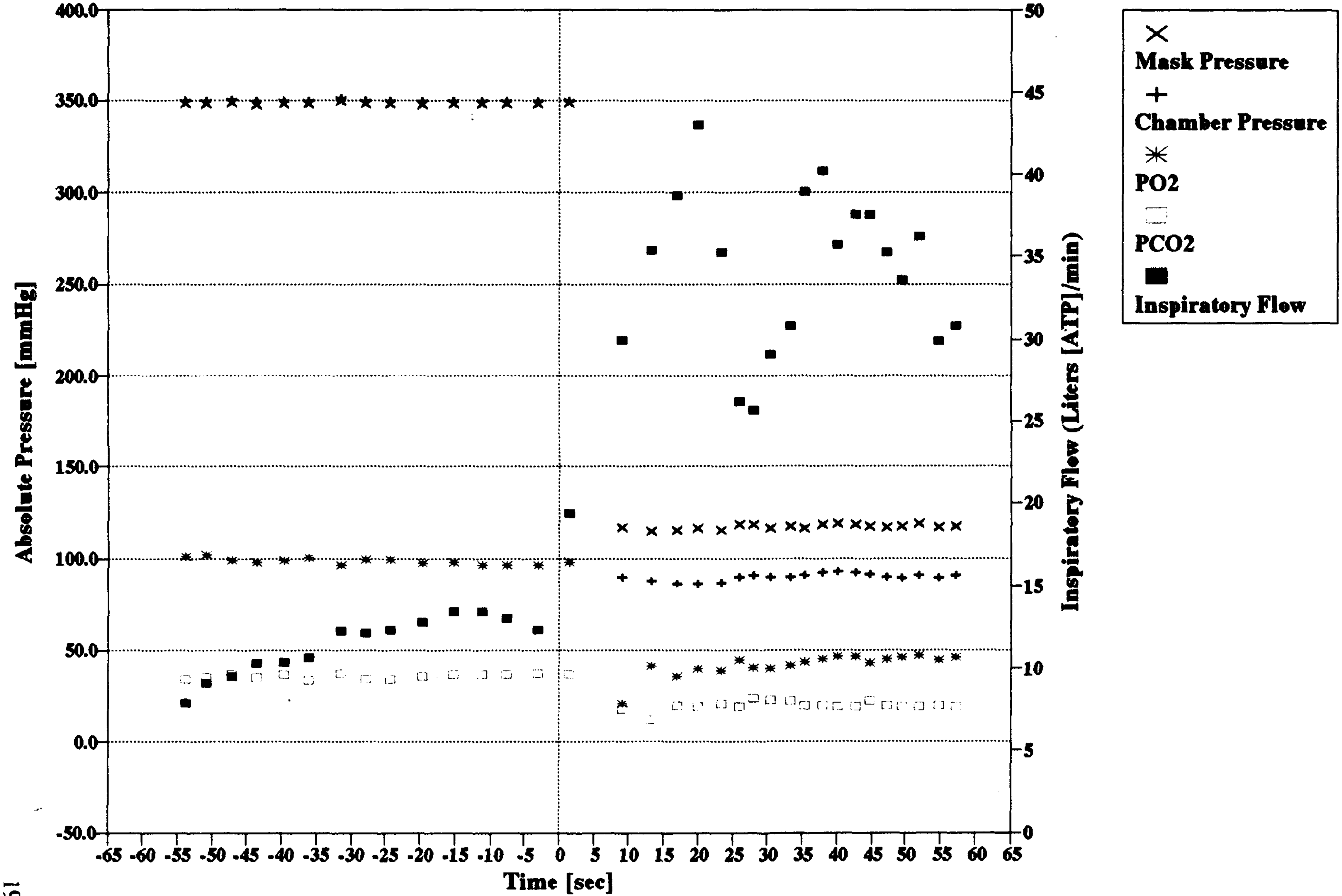


Subject: DOA / 93% O2 Dilution
20/50 kft Rapid Decompression
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65

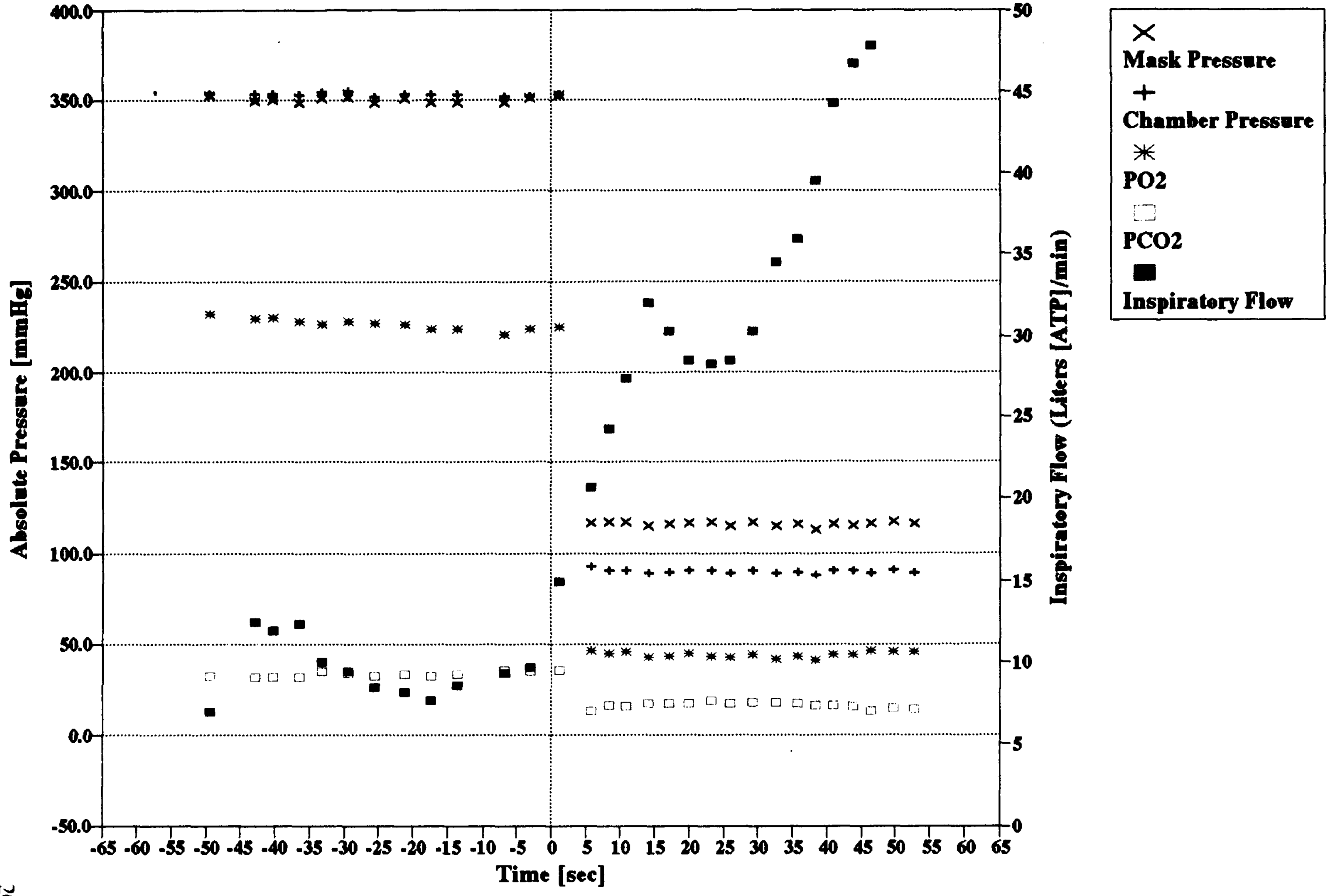
Subject: DOA / 90% O2 Non-Dilution
20/50 kft Rapid Decompression



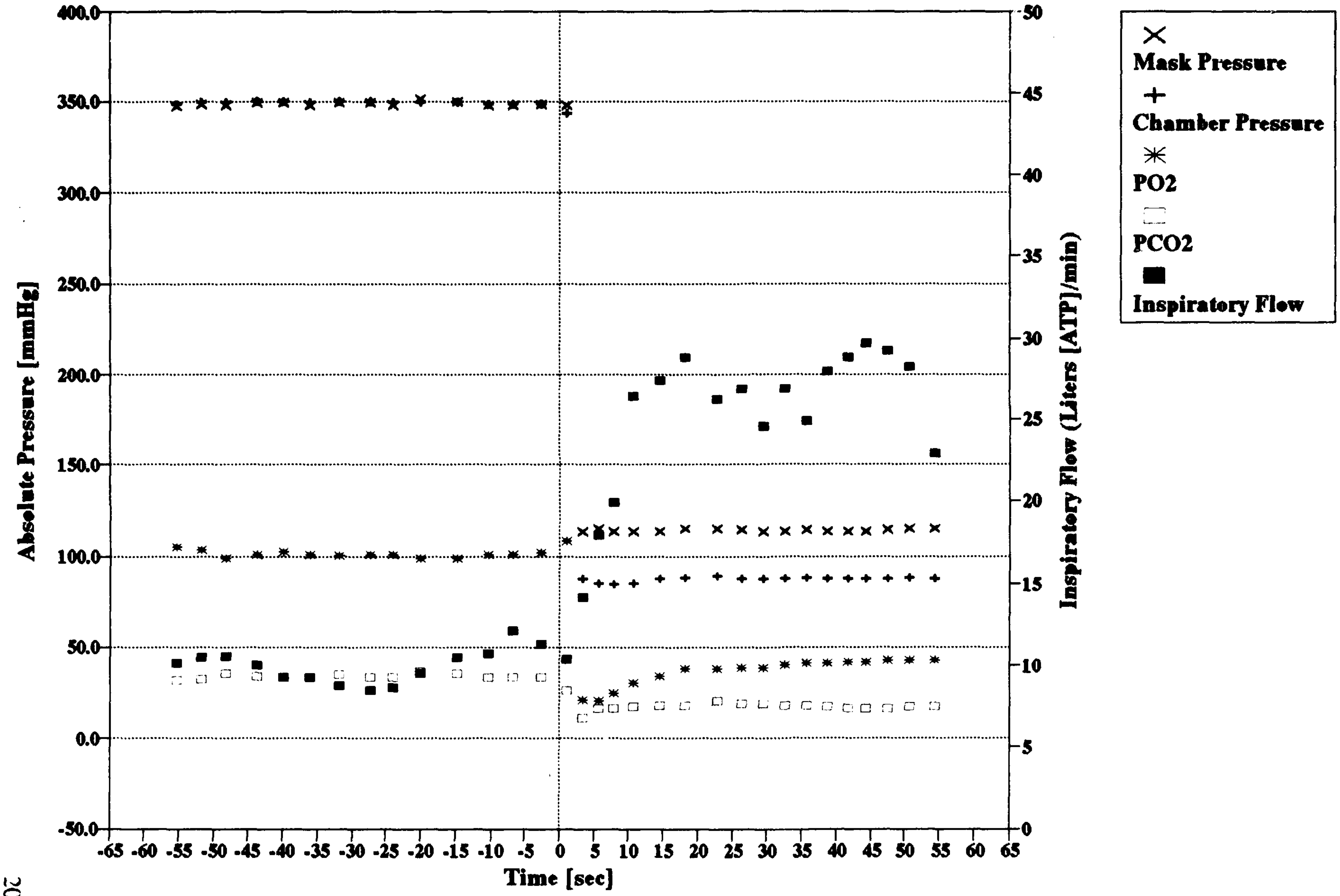
Subject: DOA / 90% O2 Dilution
20/50 kft Rapid Decompression



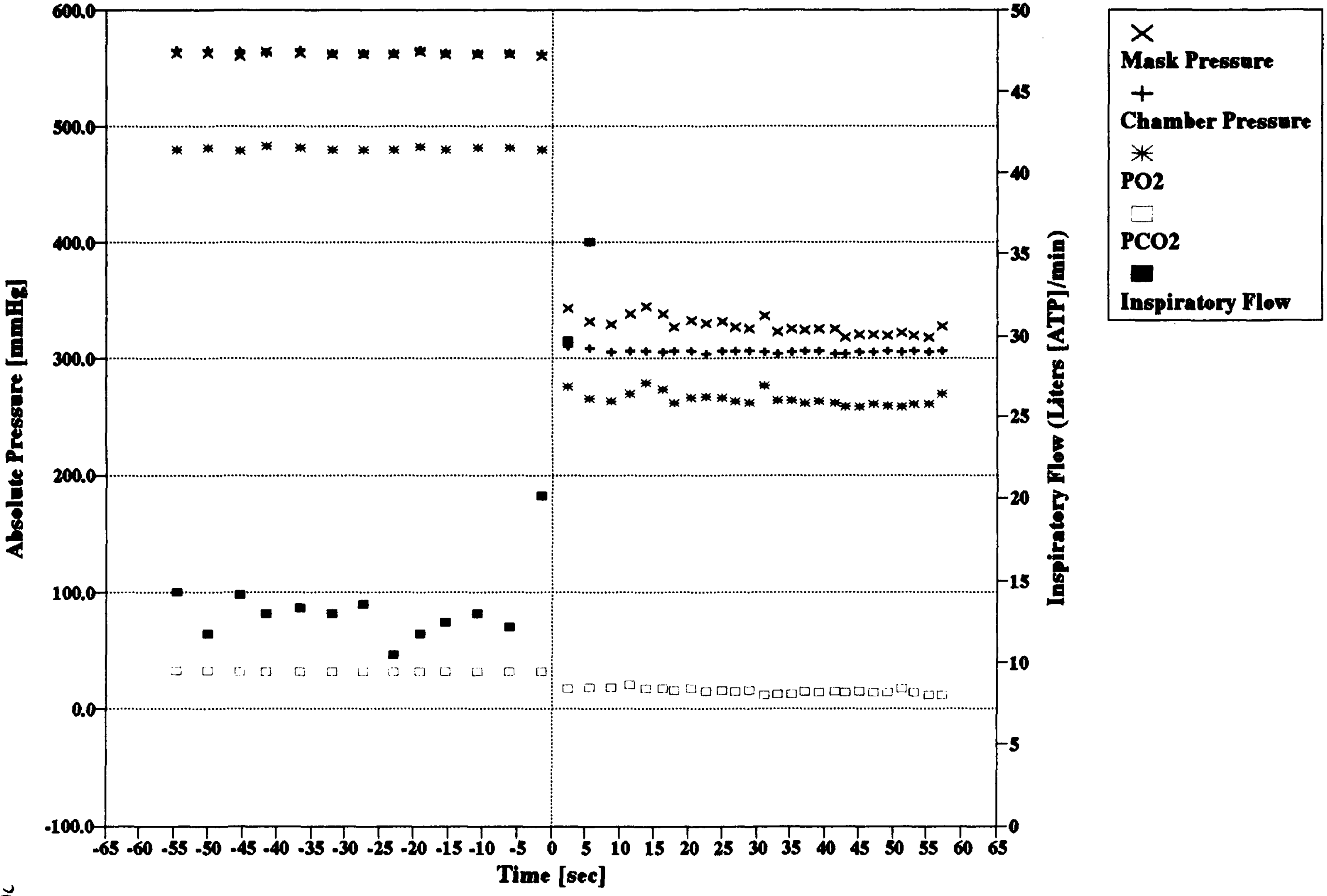
Subject: DOA / 85% O2 Non-Dilution
20/50 kft Rapid Decompression



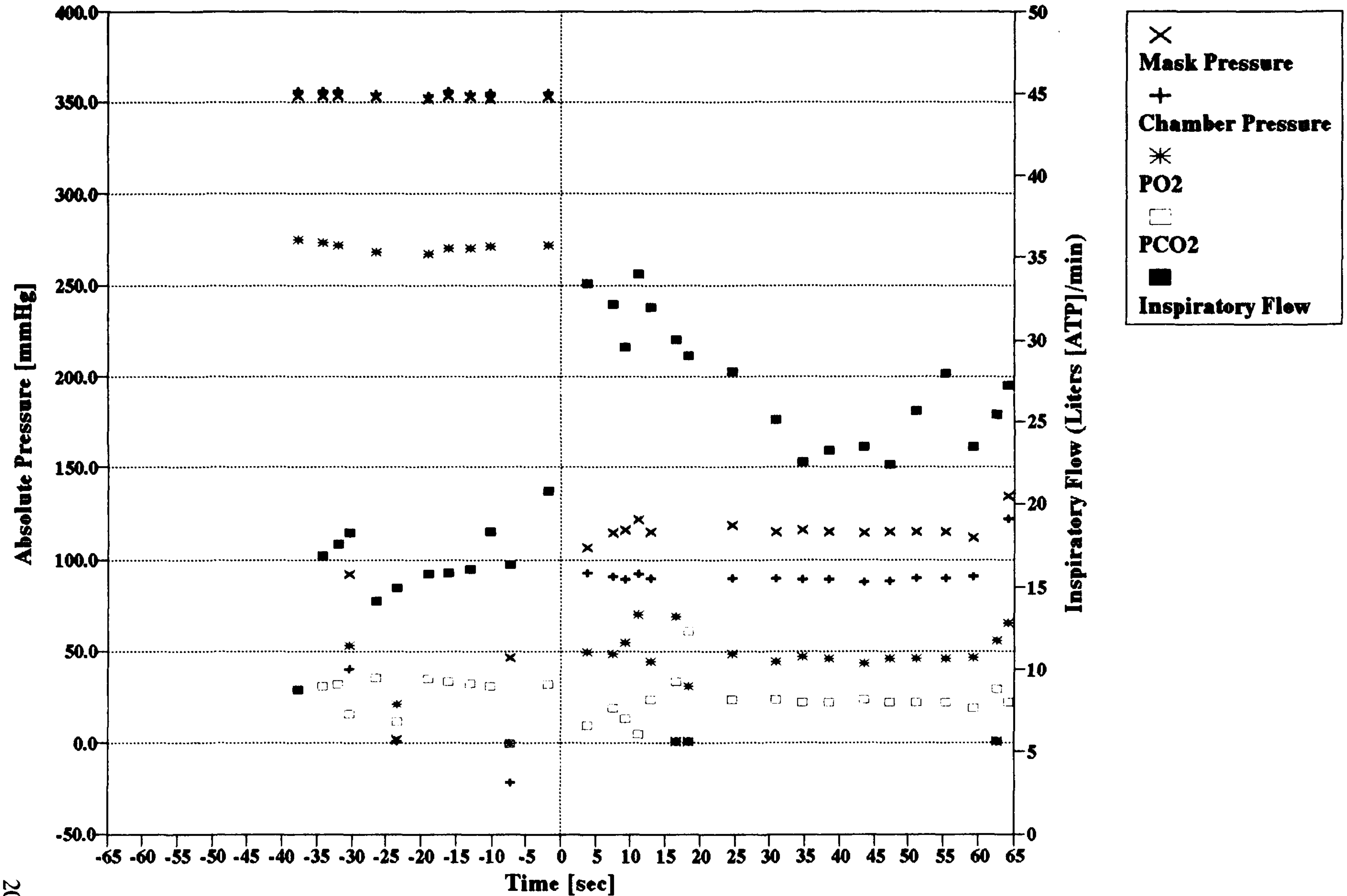
Subject: DOA / 85% O2 Dilution
20/50 kft Rapid Decompression



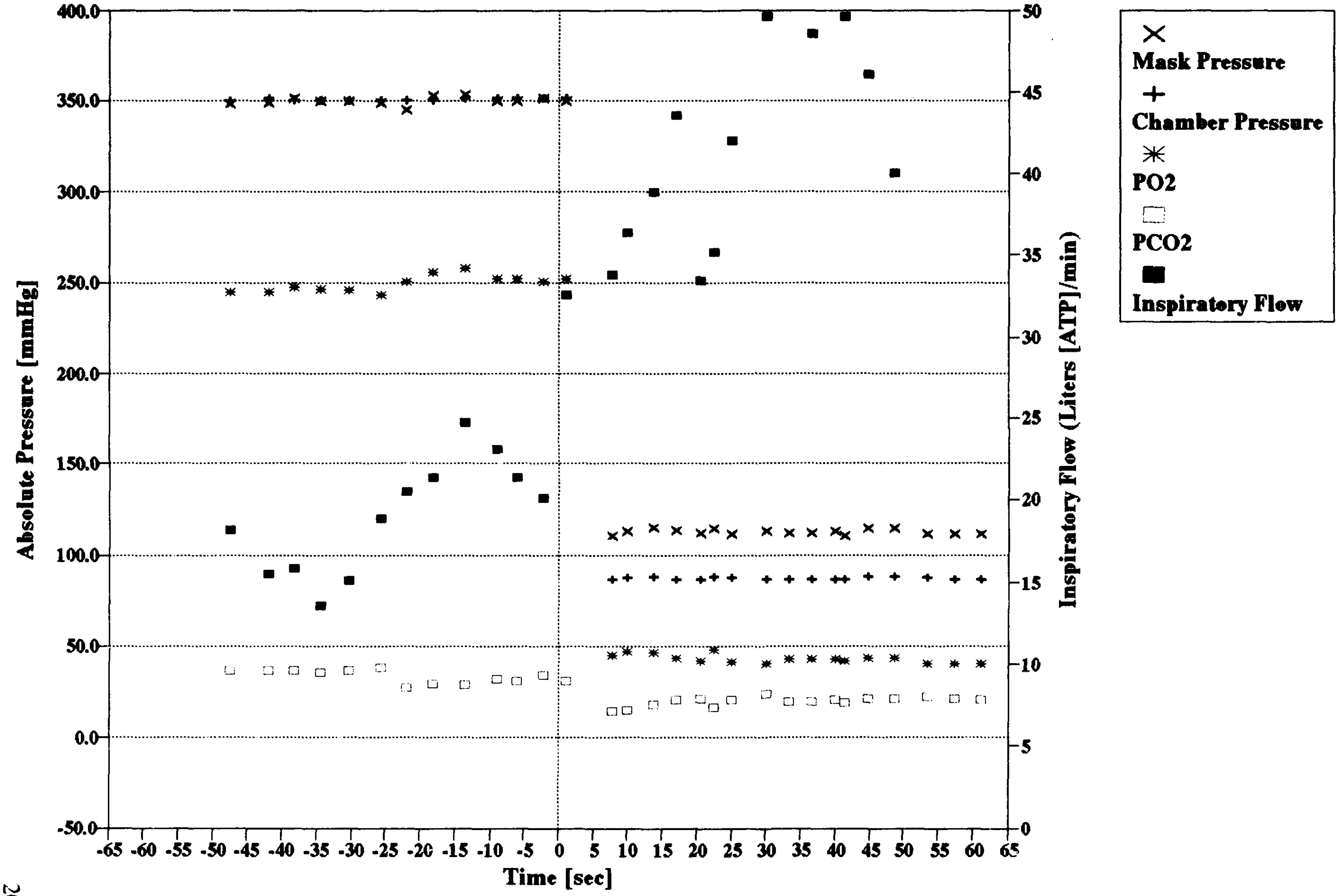
Subject: DOA / 100% O2 Non-Dilution
8/20 kft Rapid Decompression

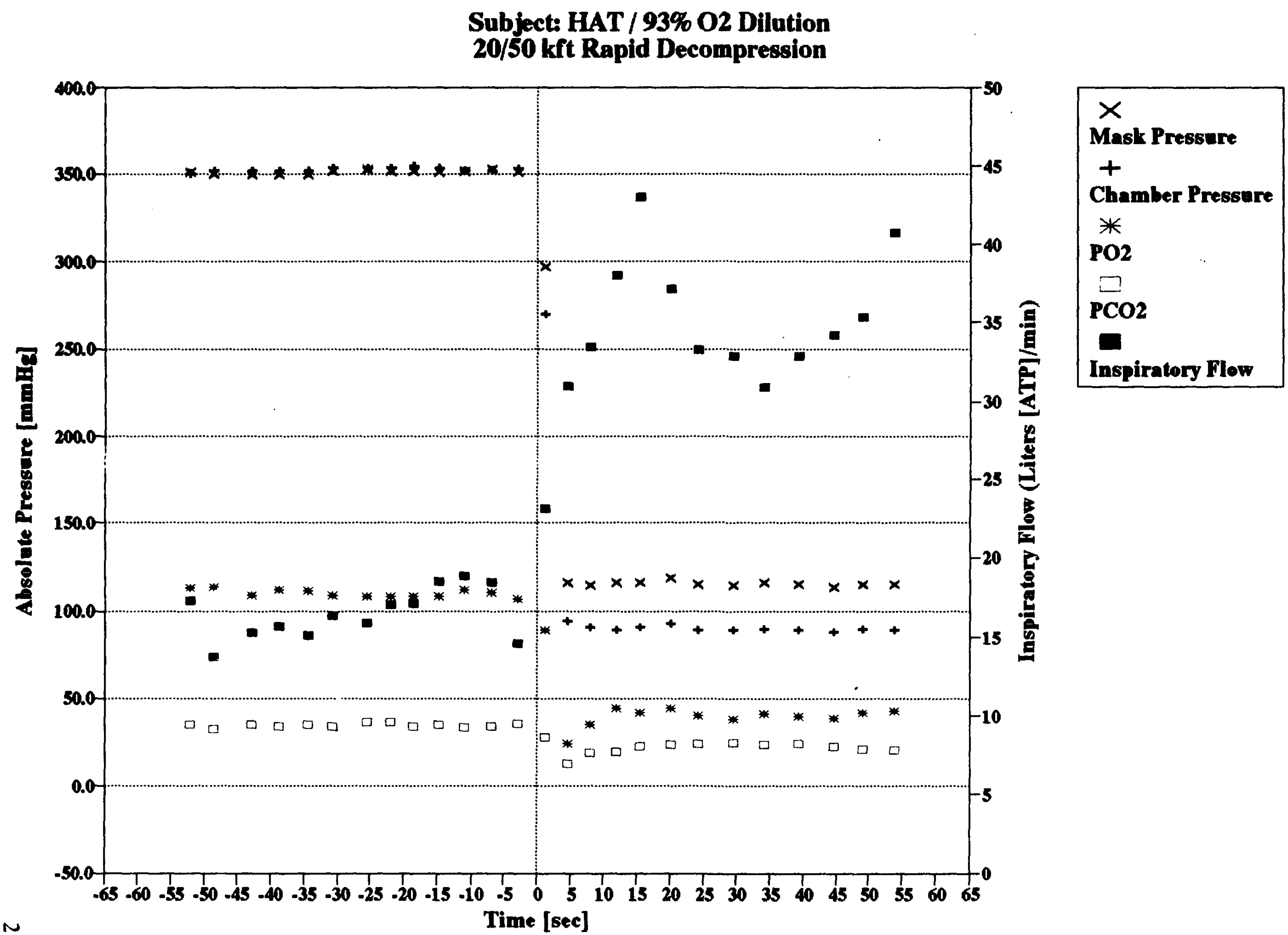


Subject: HAT / 100% O2 Non-Dilution
20/50 kft Rapid Decompression



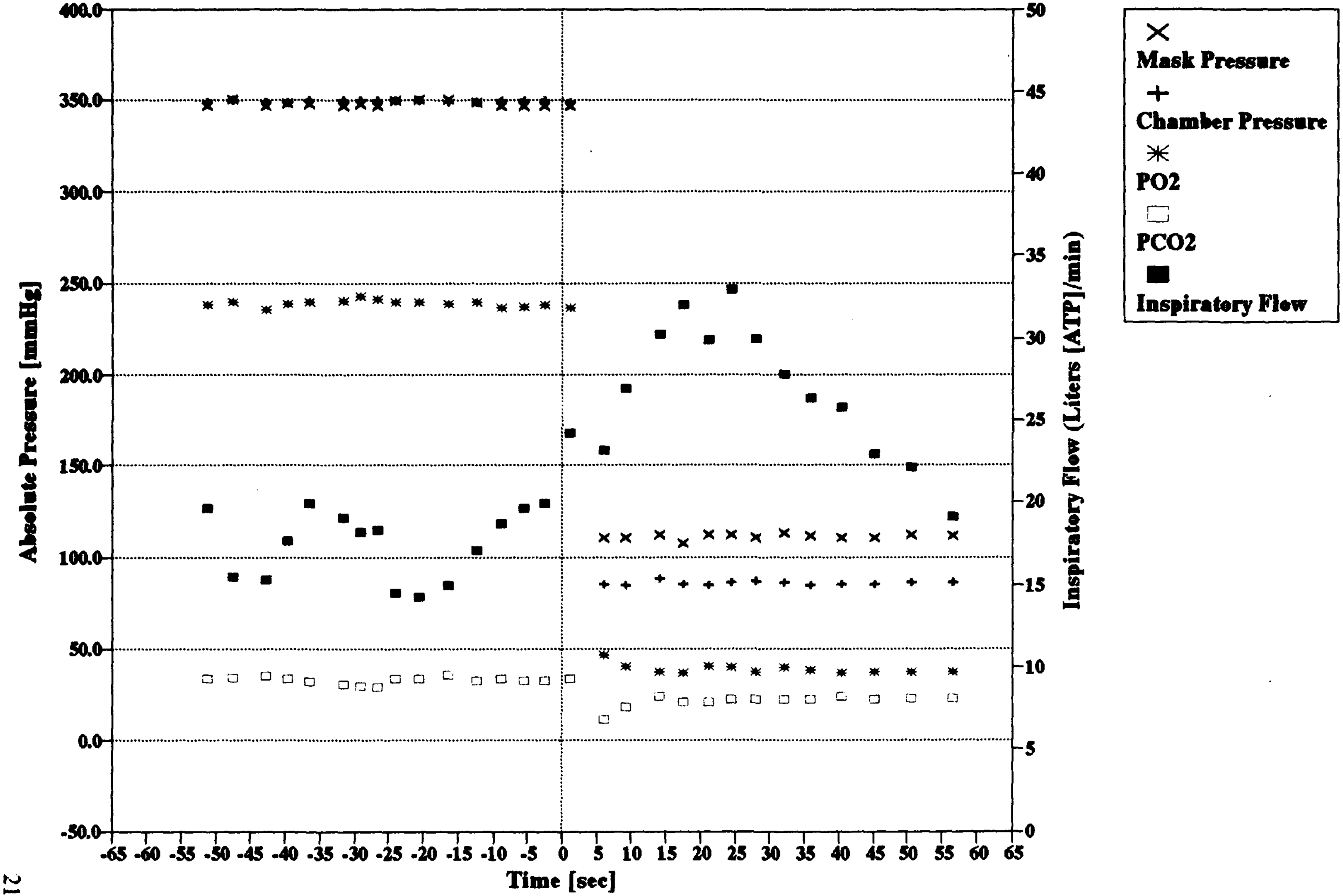
Subject: HAT / 93% O2 Non-Dilution
20/50 kft Rapid Decompression



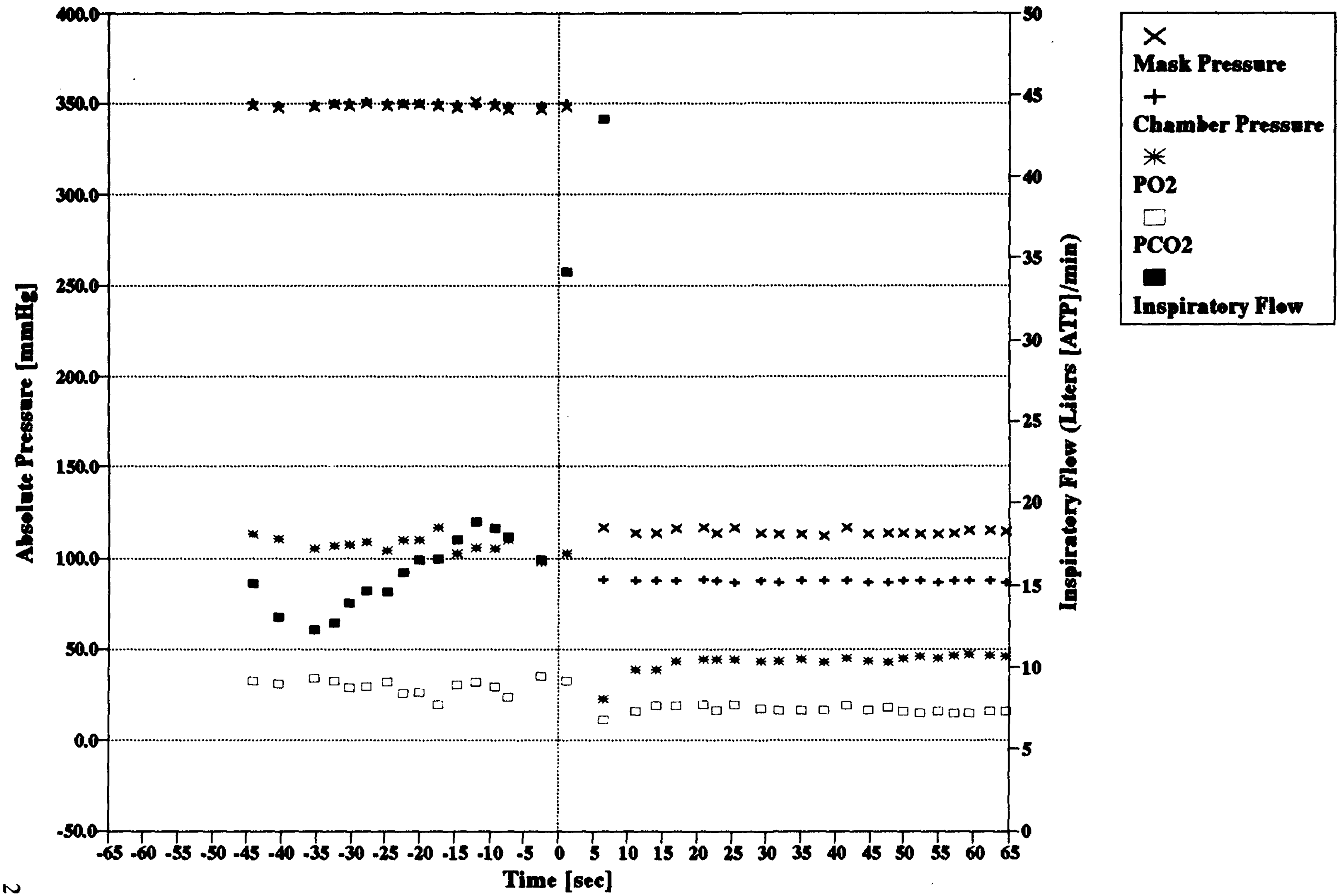


Subject: HAT / 93% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

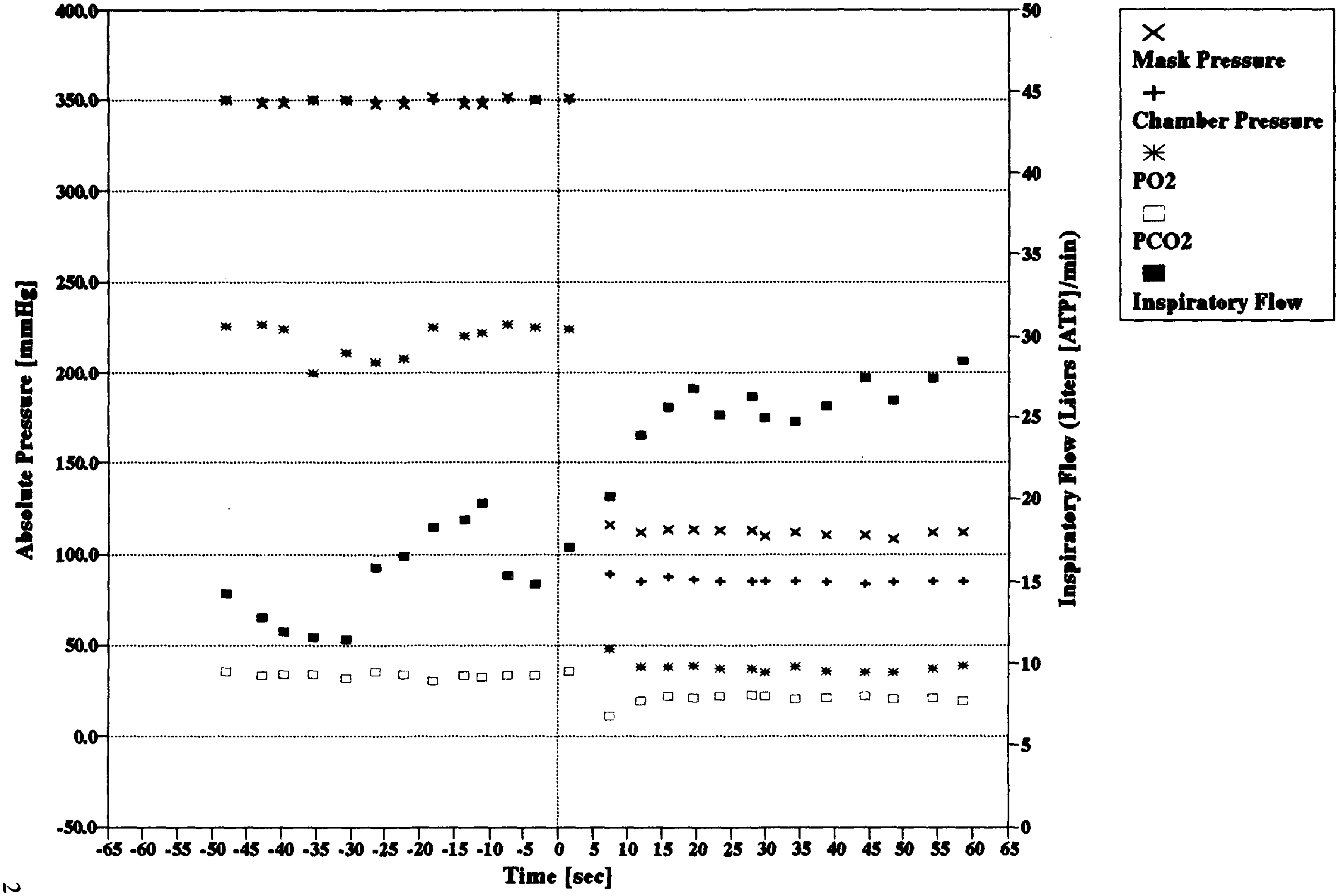
Subject: HAT / 90% O2 Non-Dilution
20/50 kft Rapid Decompression



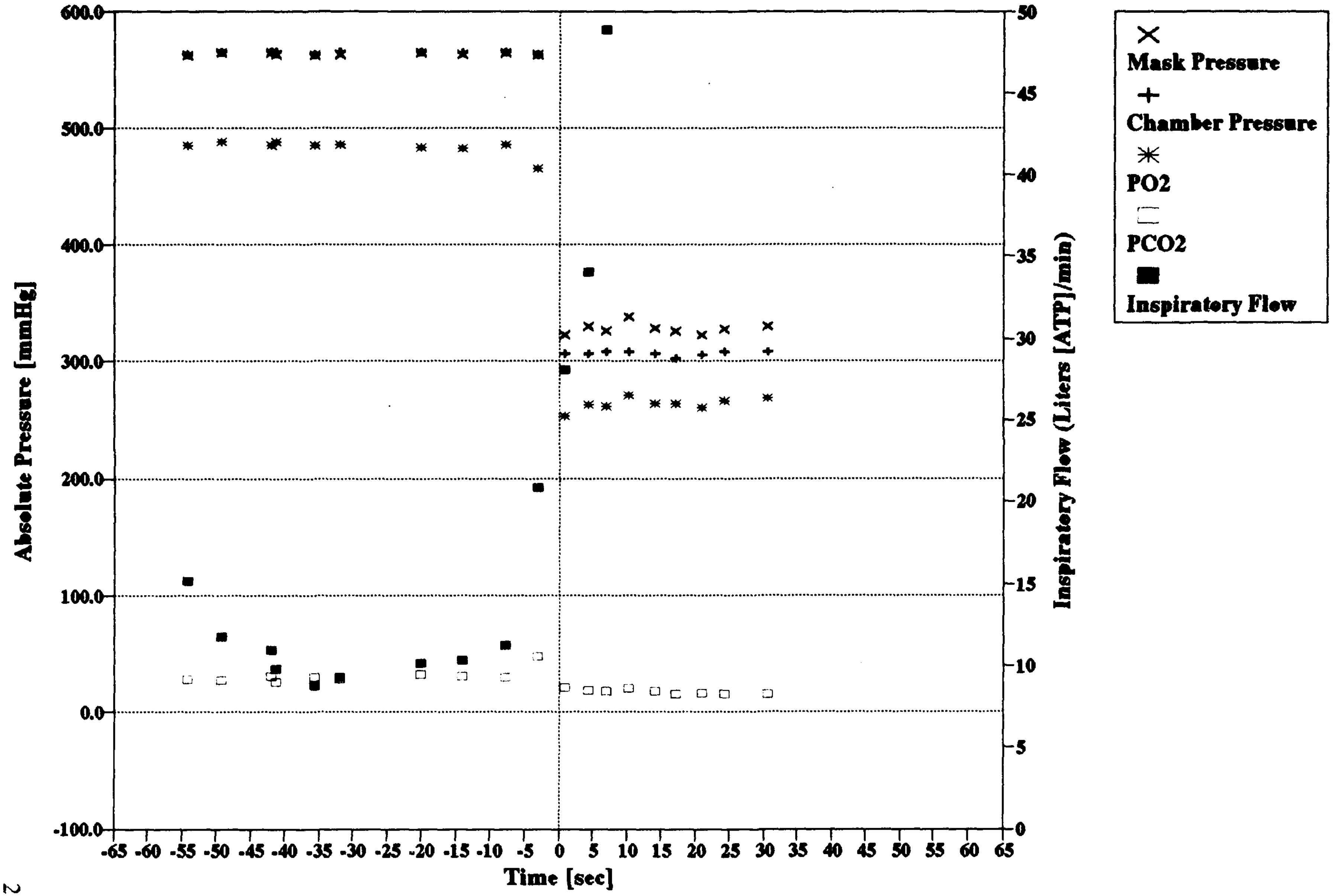
Subject: HAT / 90% O2 Dilution
20/50 kft Rapid Decompression

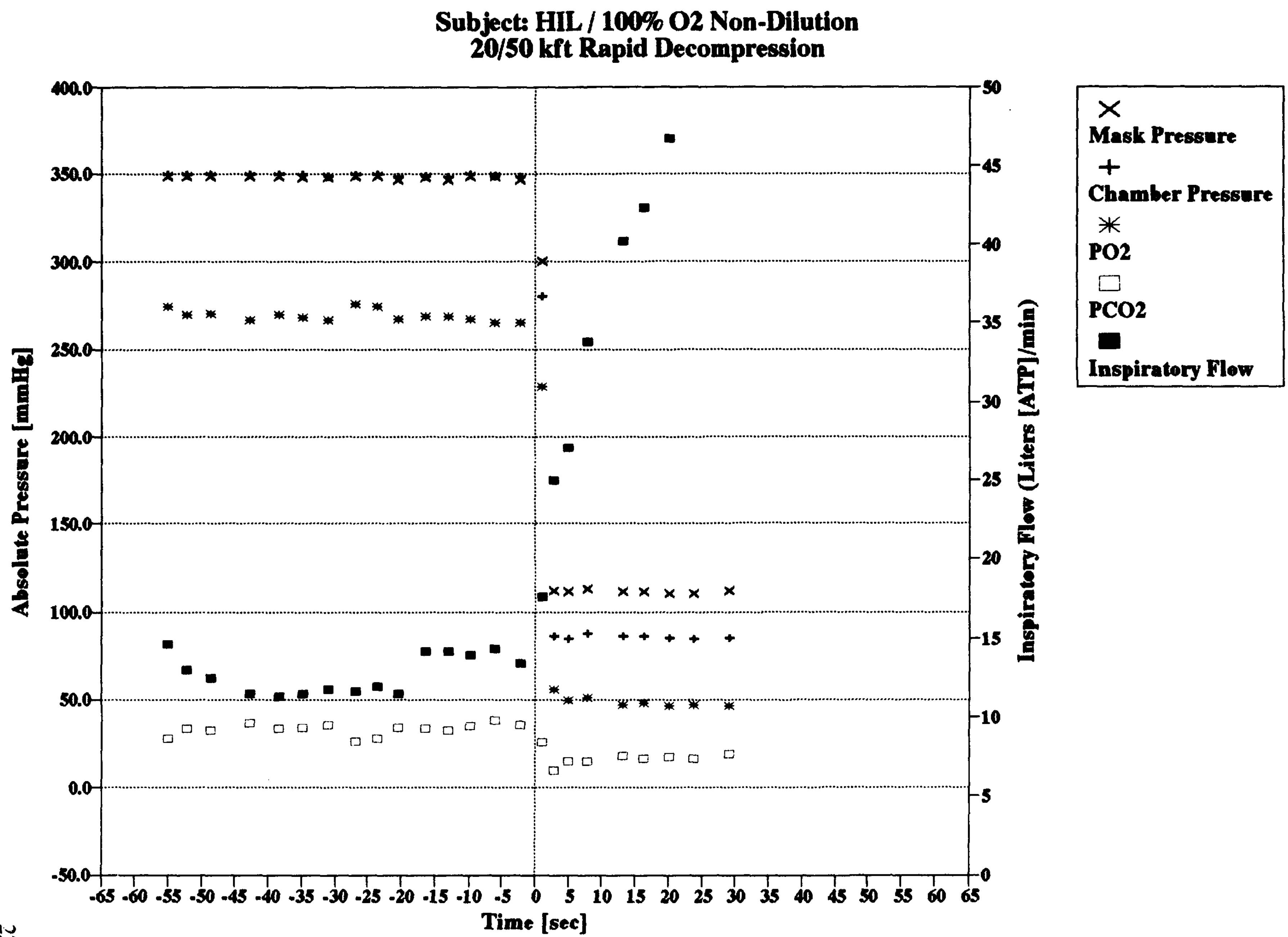


Subject: HAT / 85% O2 Non-Dilution
20/50 kft Rapid Decompression



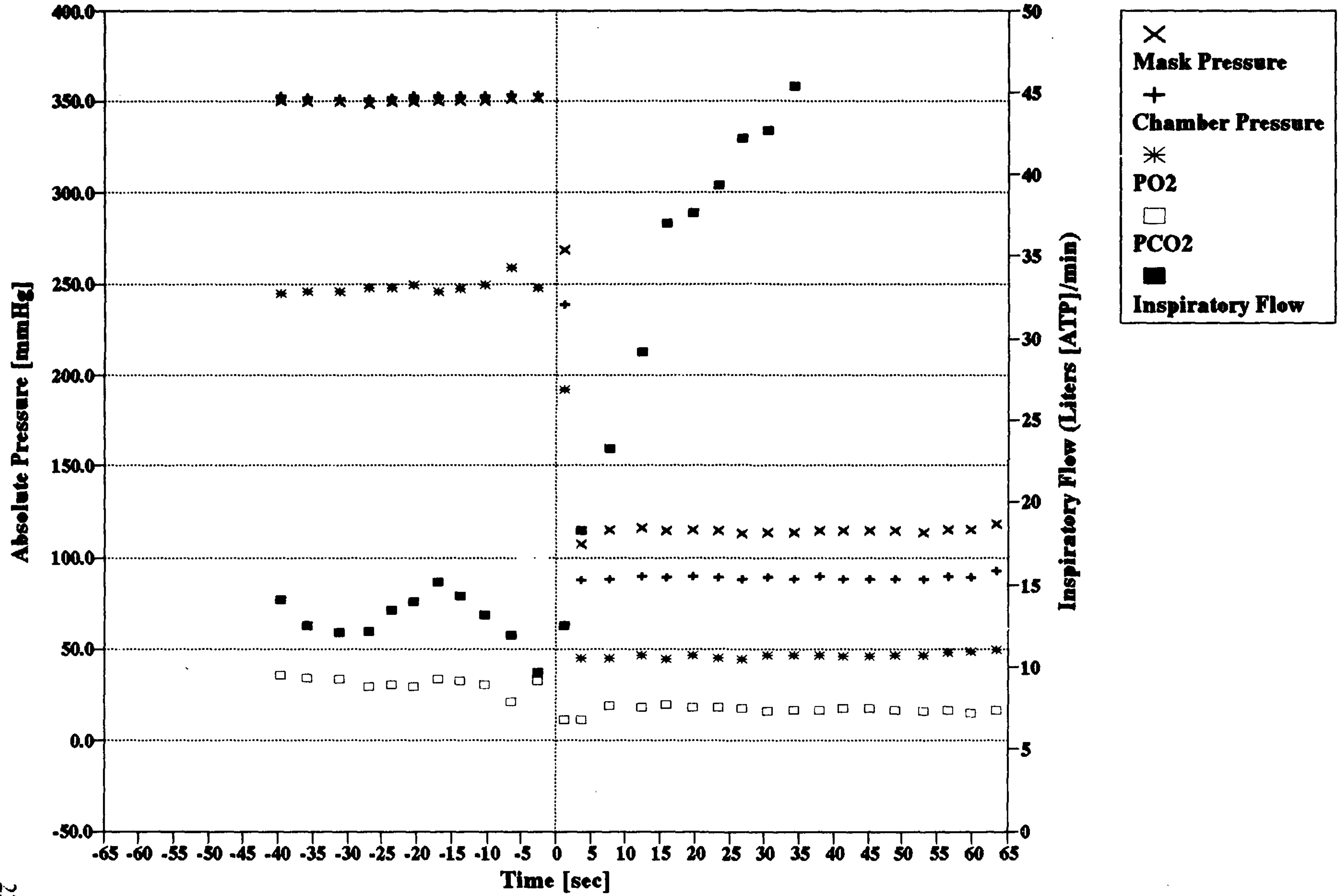
Subject: HAT / 100% O2 Non-Dilution
8/20 kft Rapid Decompression

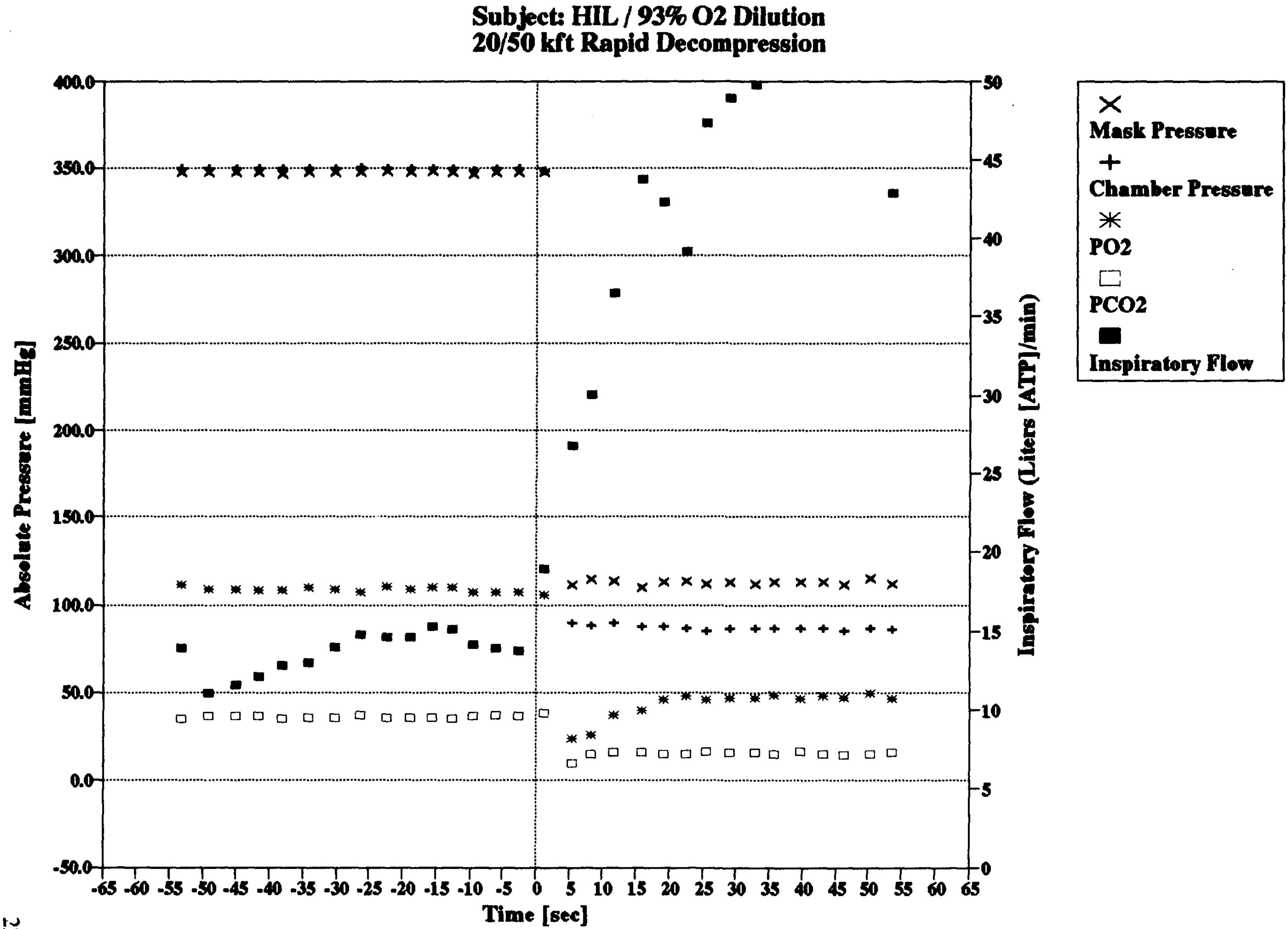




Subject: HIL / 100% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

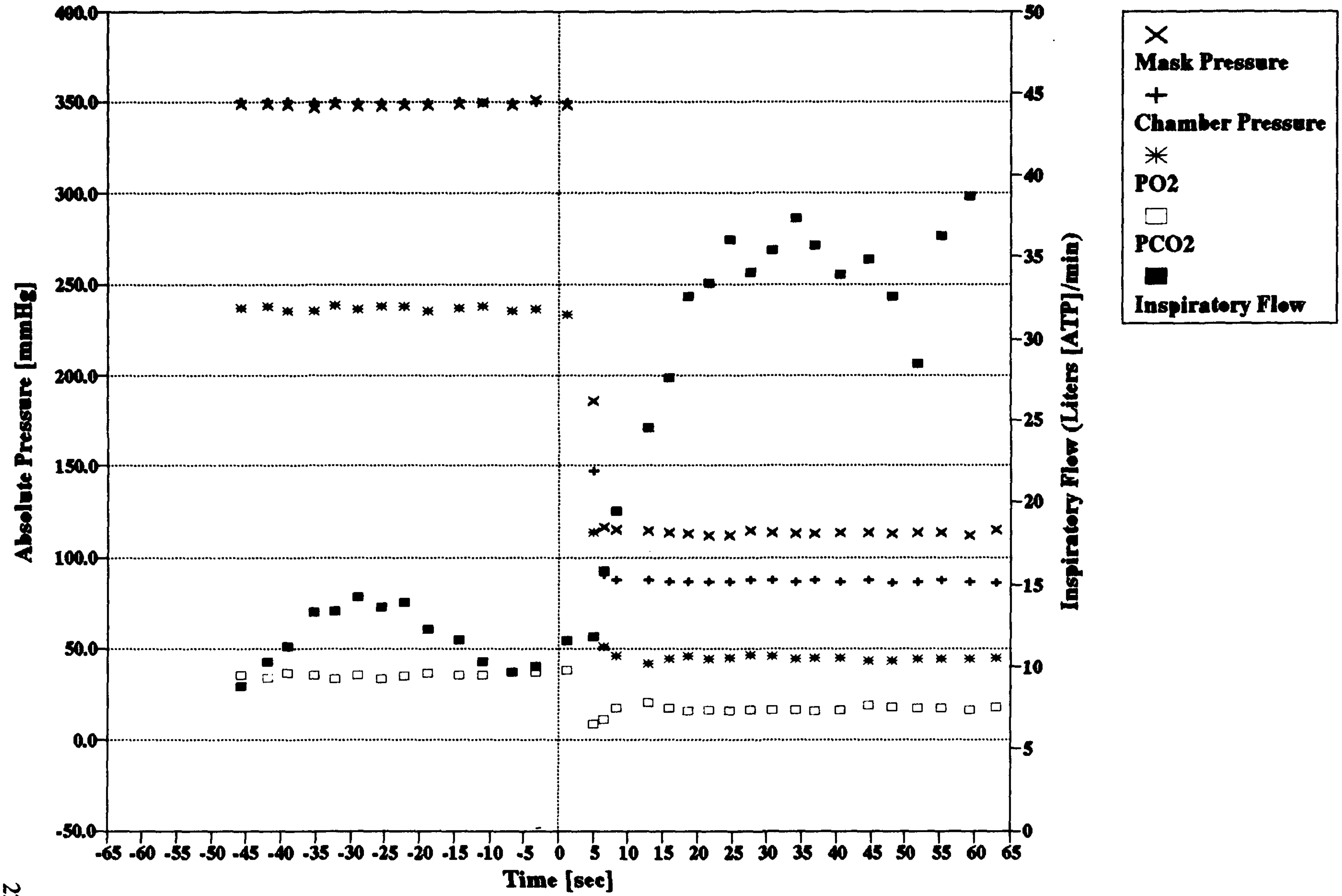
Subject: HIL / 93% O2 Non-Dilution
20/50 kft Rapid Decompression



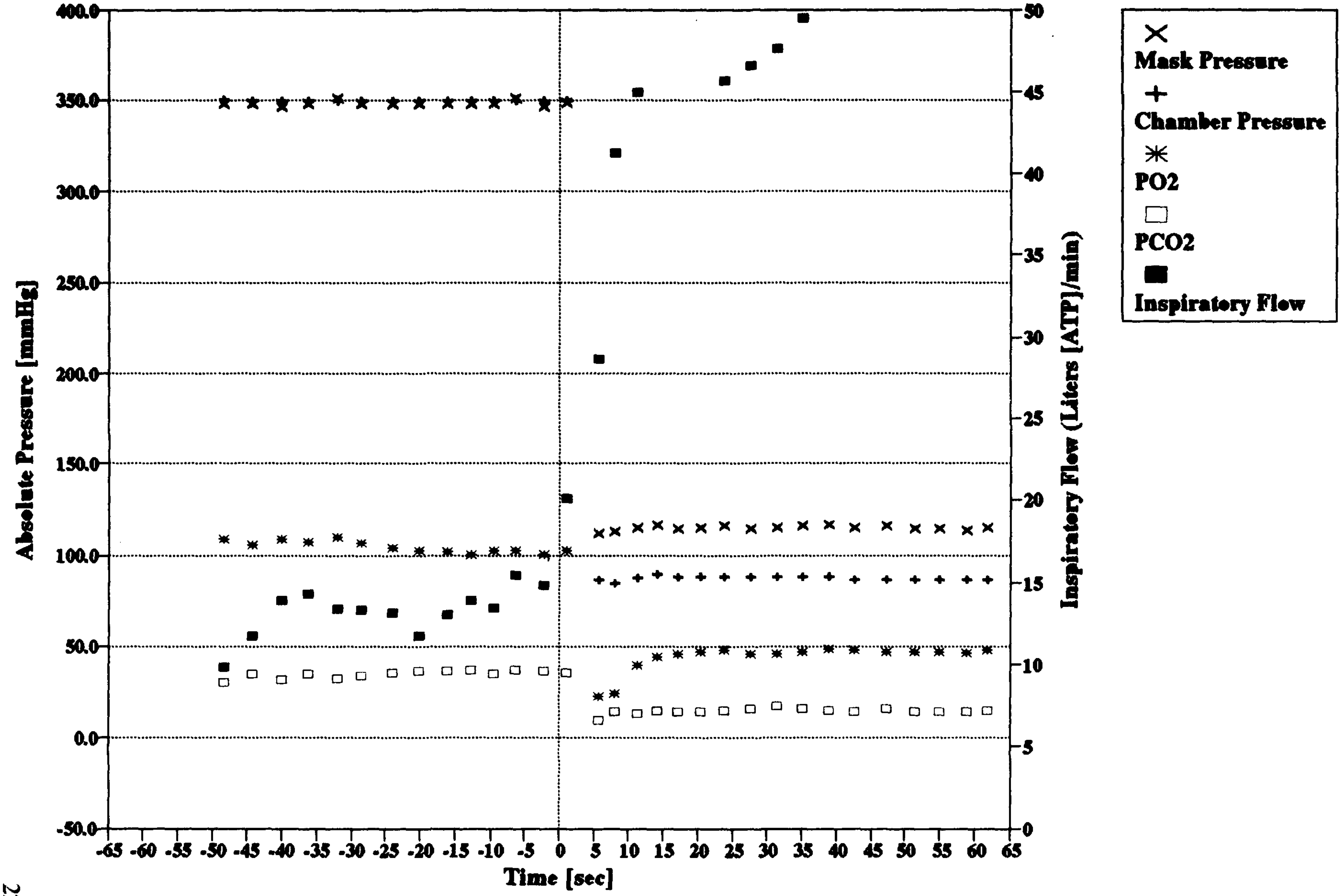


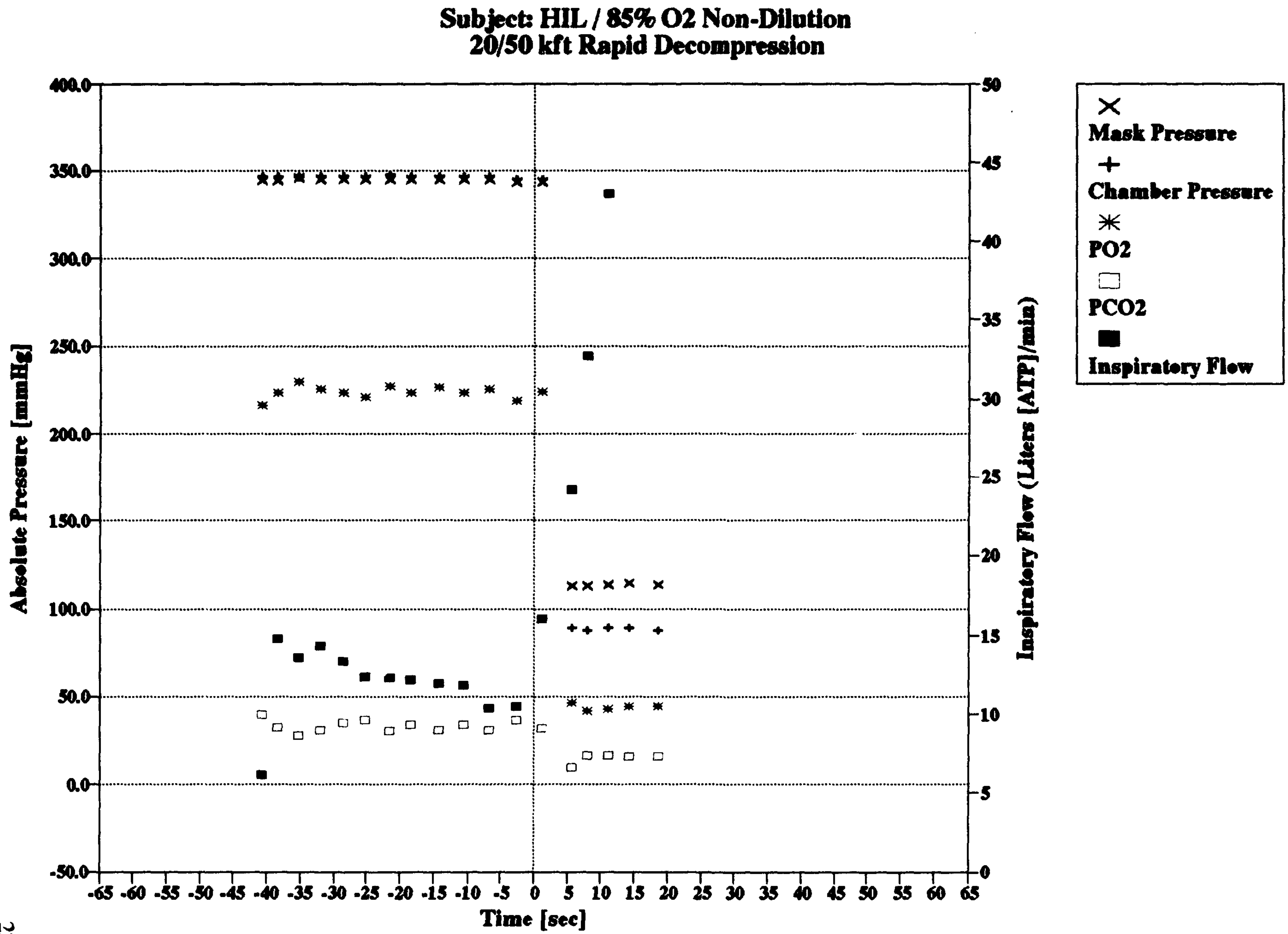
Subject: HIL / 93% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: HIL / 90% O2 Non-Dilution
20/50 kft Rapid Decompression

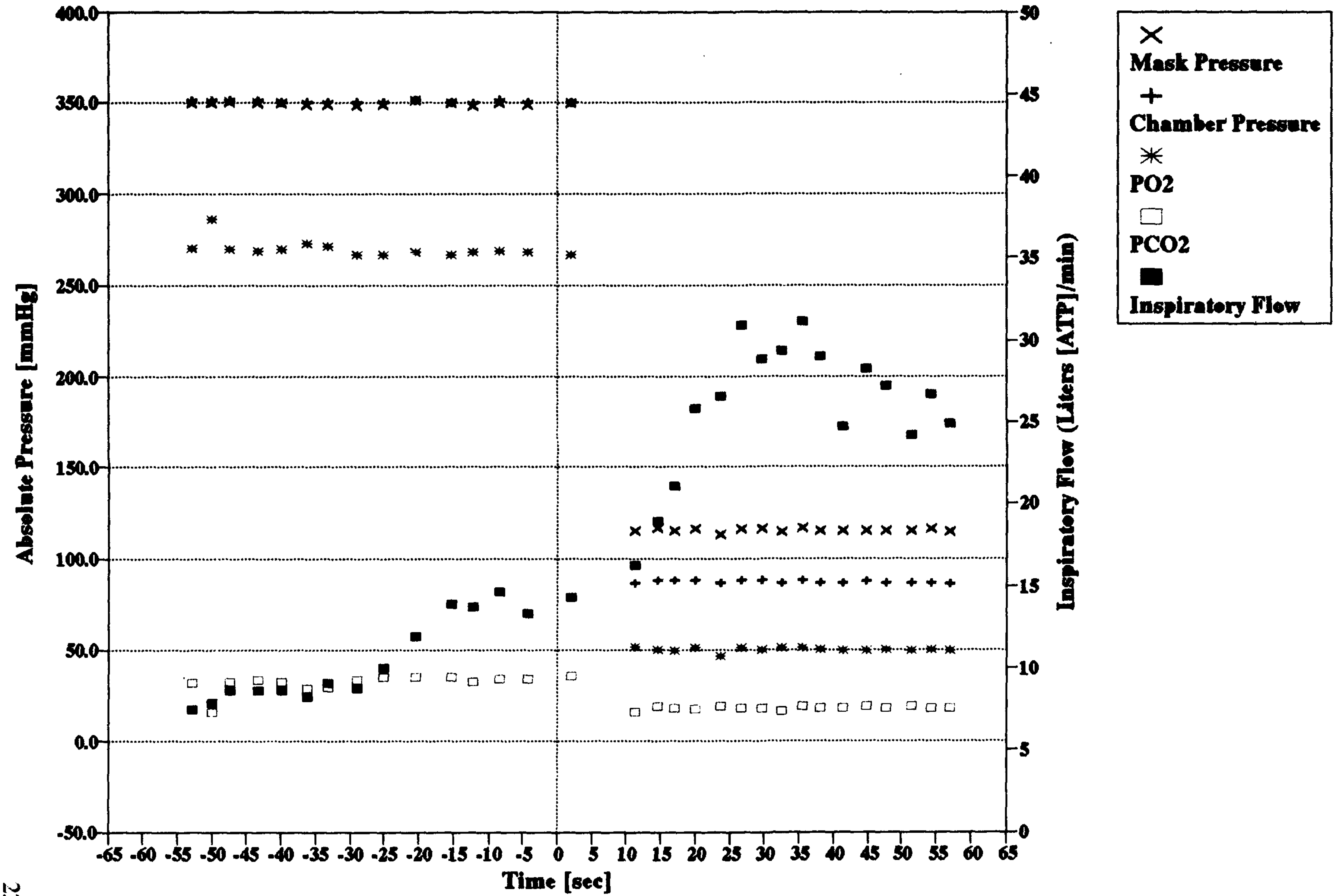


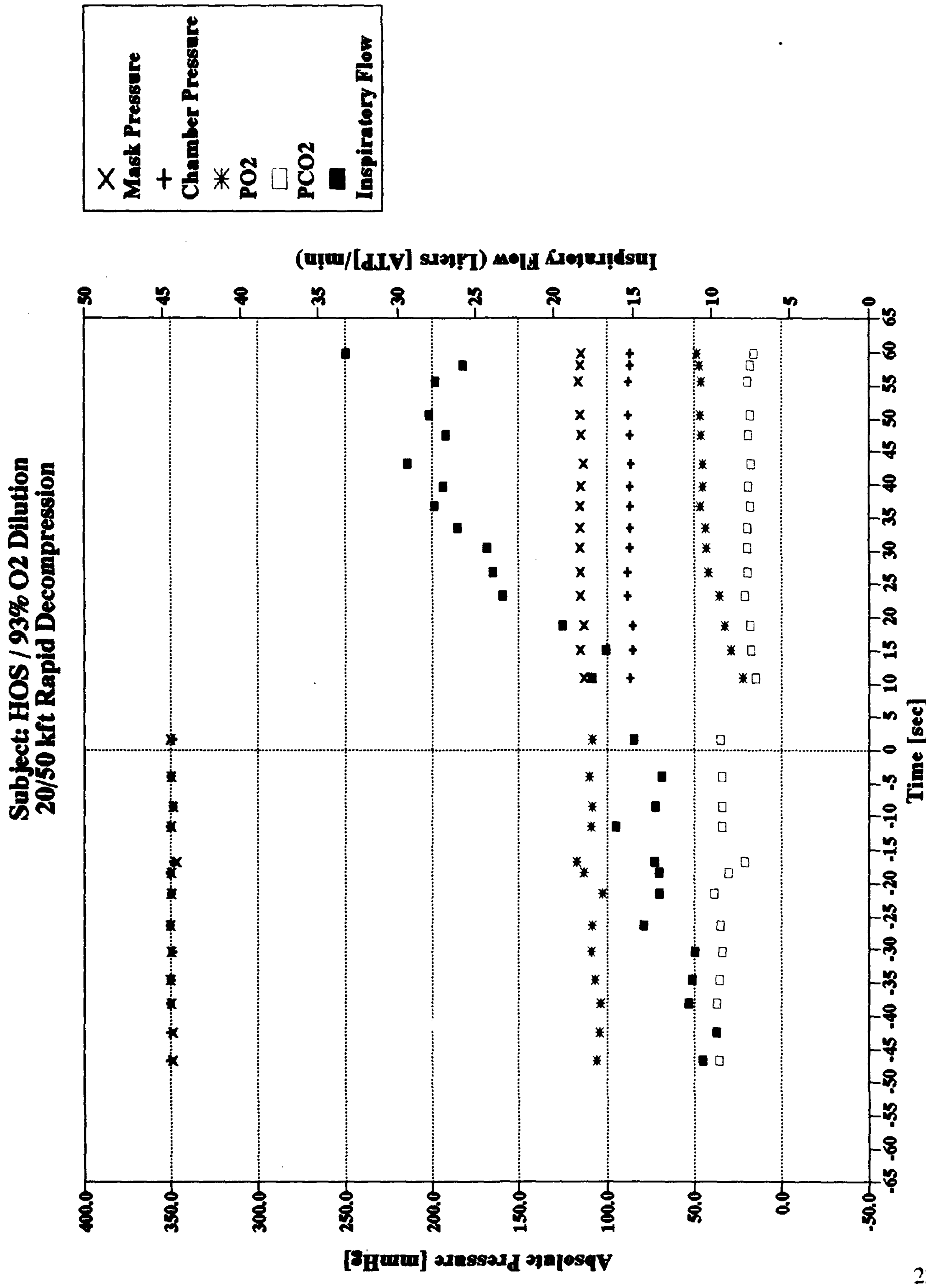
Subject: HIL / 90% O2 Dilution
20/50 kft Rapid Decompression

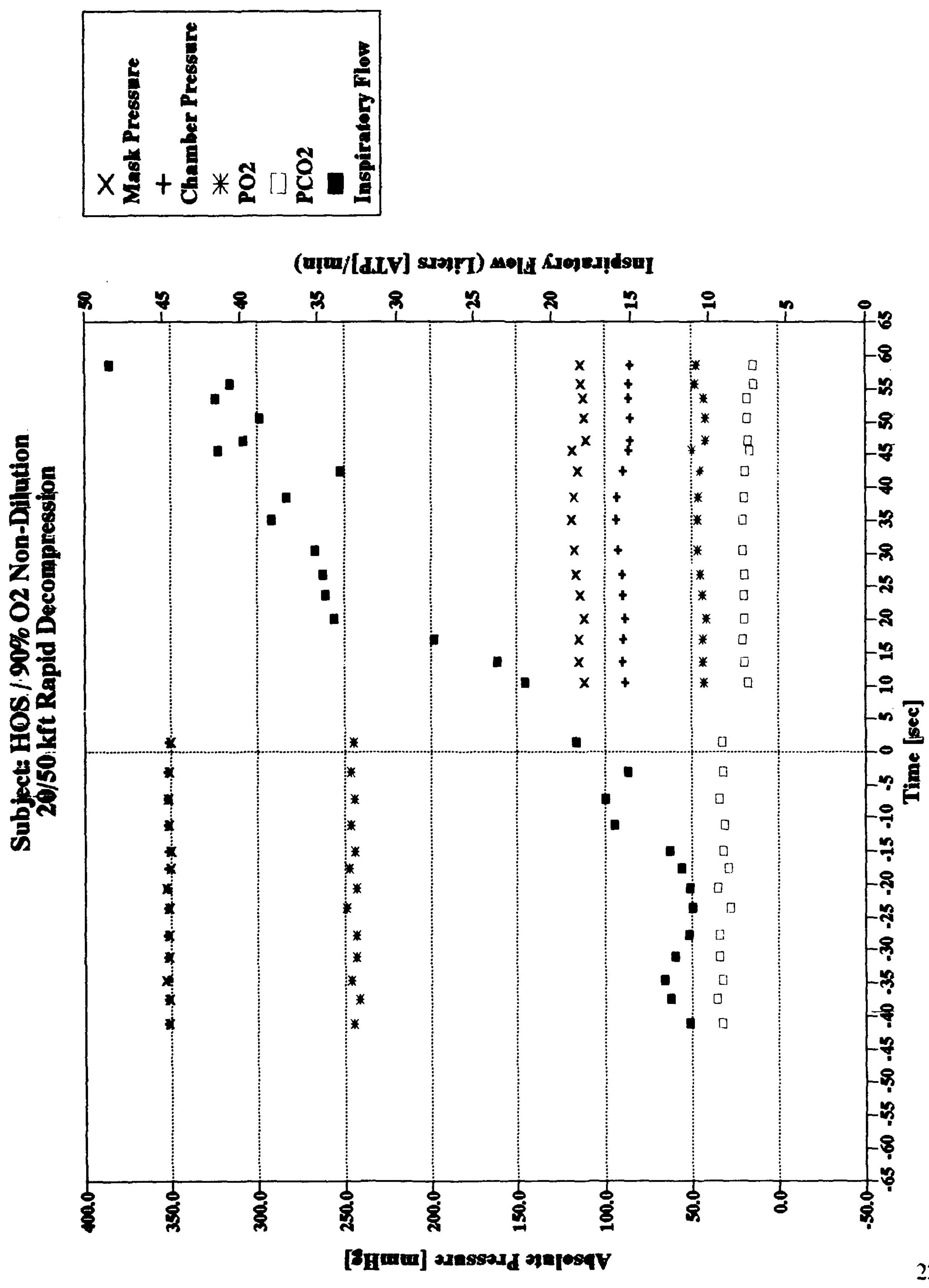



Subject: HIL / 85% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: HOS / 100% O2 Non-Dilution
20/50 kft Rapid Decompression




Subject: HOS / 93% O2 Dilution
20/50 kft Rapid Decompression
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow
Inspiratory Flow (Liters [ATP]/min)
Absolute Pressure [mmHg]
Time [sec]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
50
45
40
35
30
25
20
15
10
5
0
-65
-60
-55
-50
-45
-40
-35
-30
-25
-20
-15
-10
-5
0
5
10
15
20
25
30
35
40
45
50
55
60
65


Subject: HOS / 90% O2 Non-Dilution
20/50 kft Rapid Decompression
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
50
45
40
35
30
25
20
15
10
5
0
-65
-60
-55
-50
-45
-40
-35
-30
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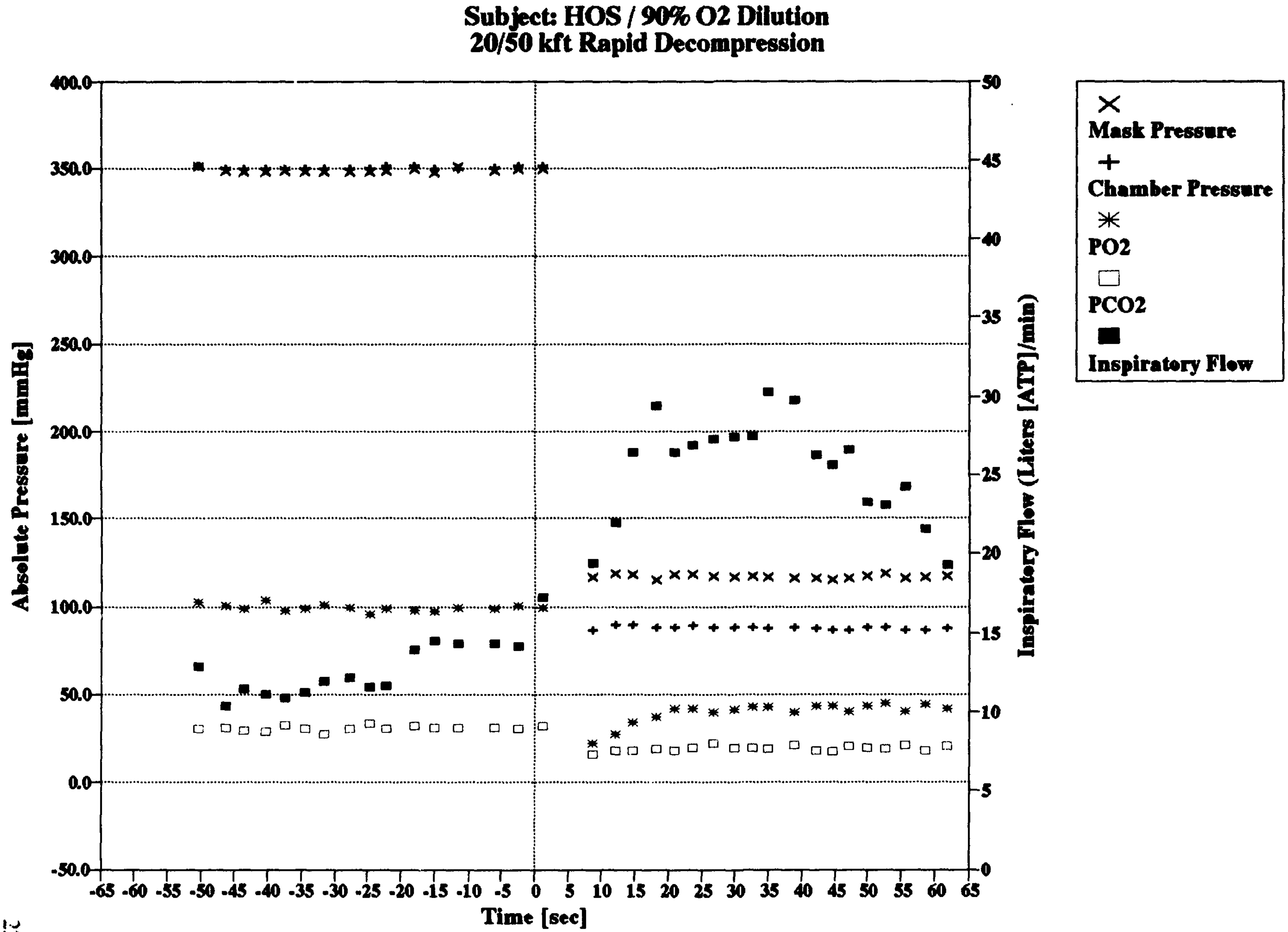
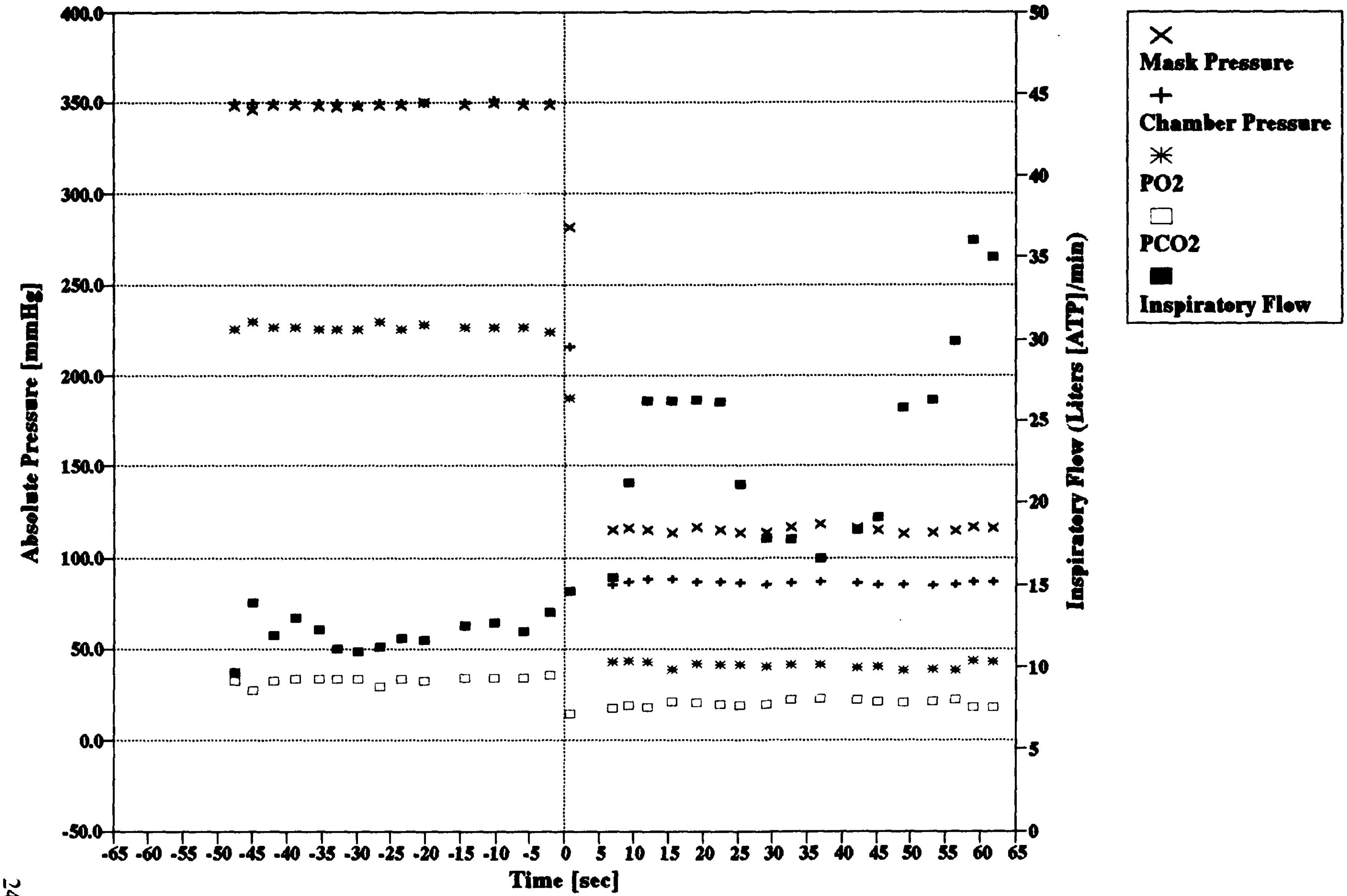
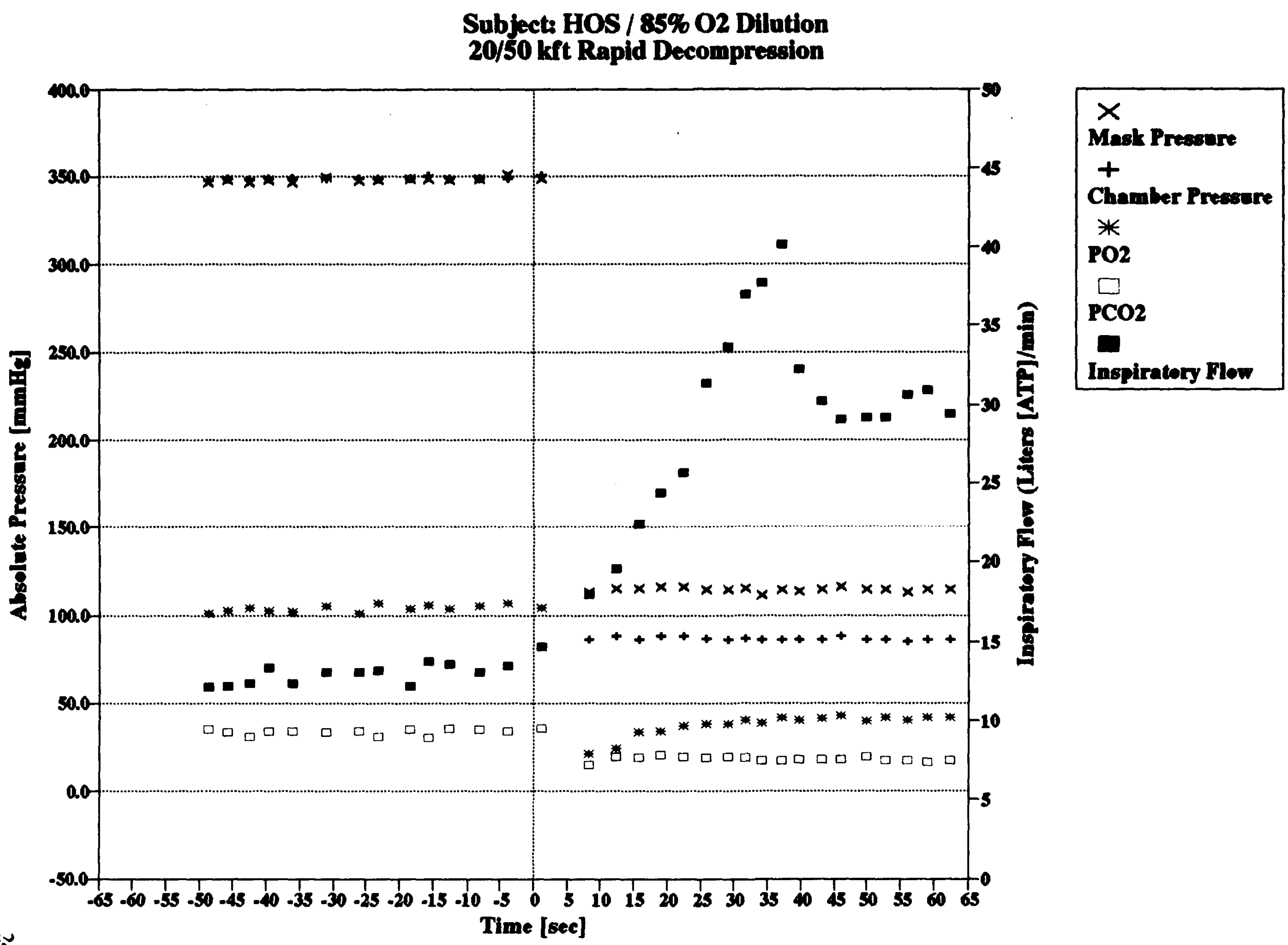
-5
0
5
10
15
20
25
30
35
40
45
50
55
60
65

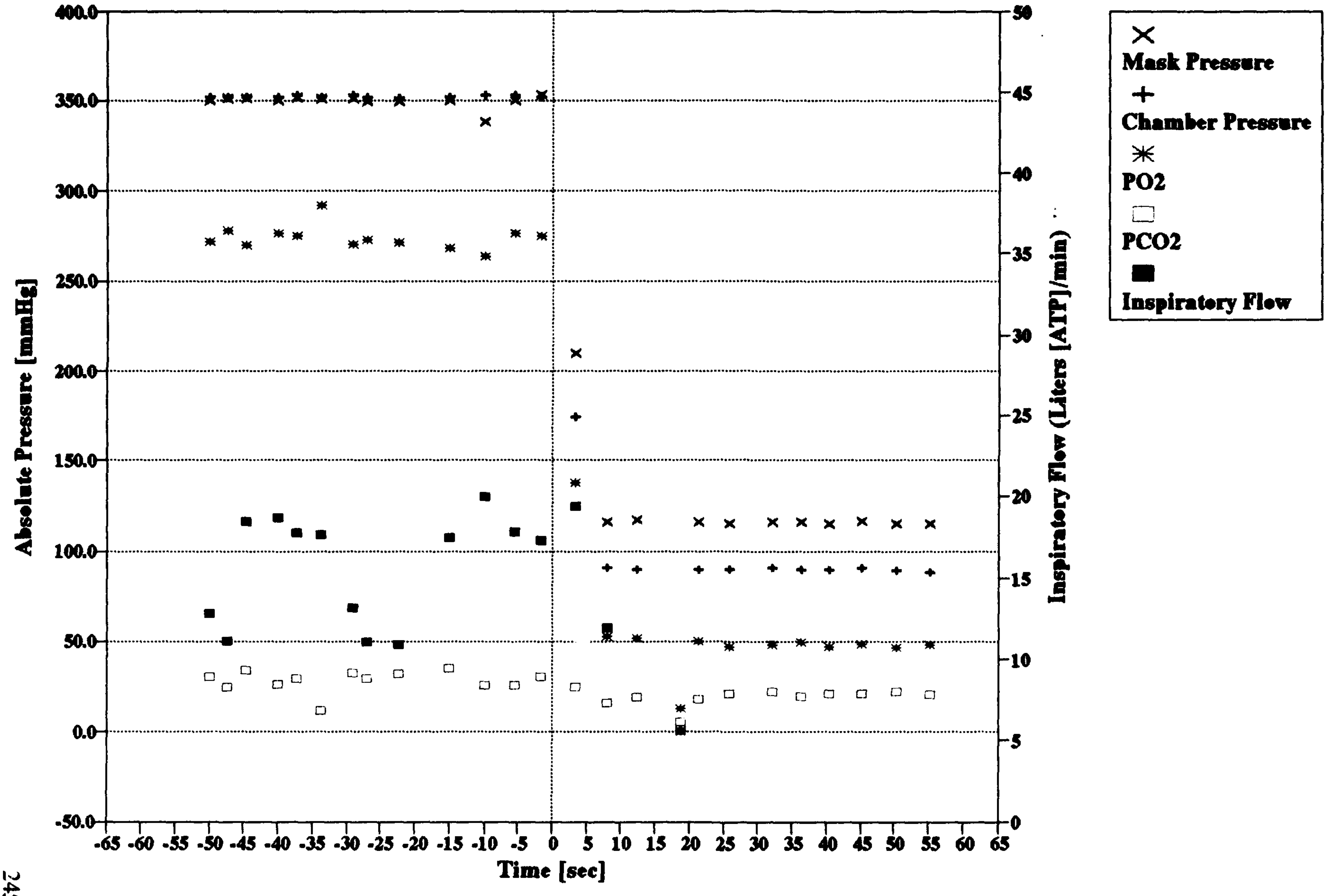

Subject: HOS / 90% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

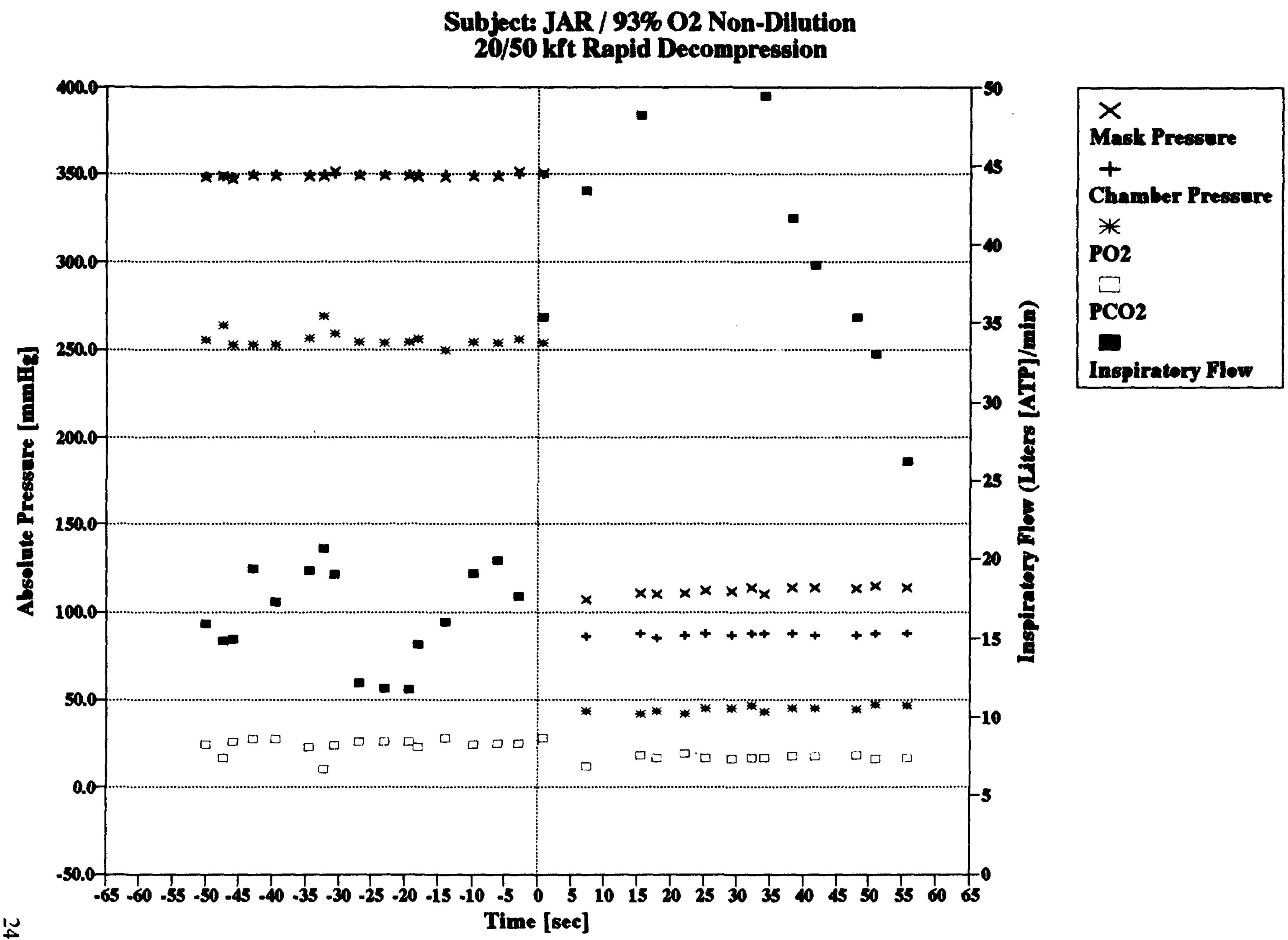
Subject: HOS / 85% O2 Non-Dilution
20/50 kft Rapid Decompression



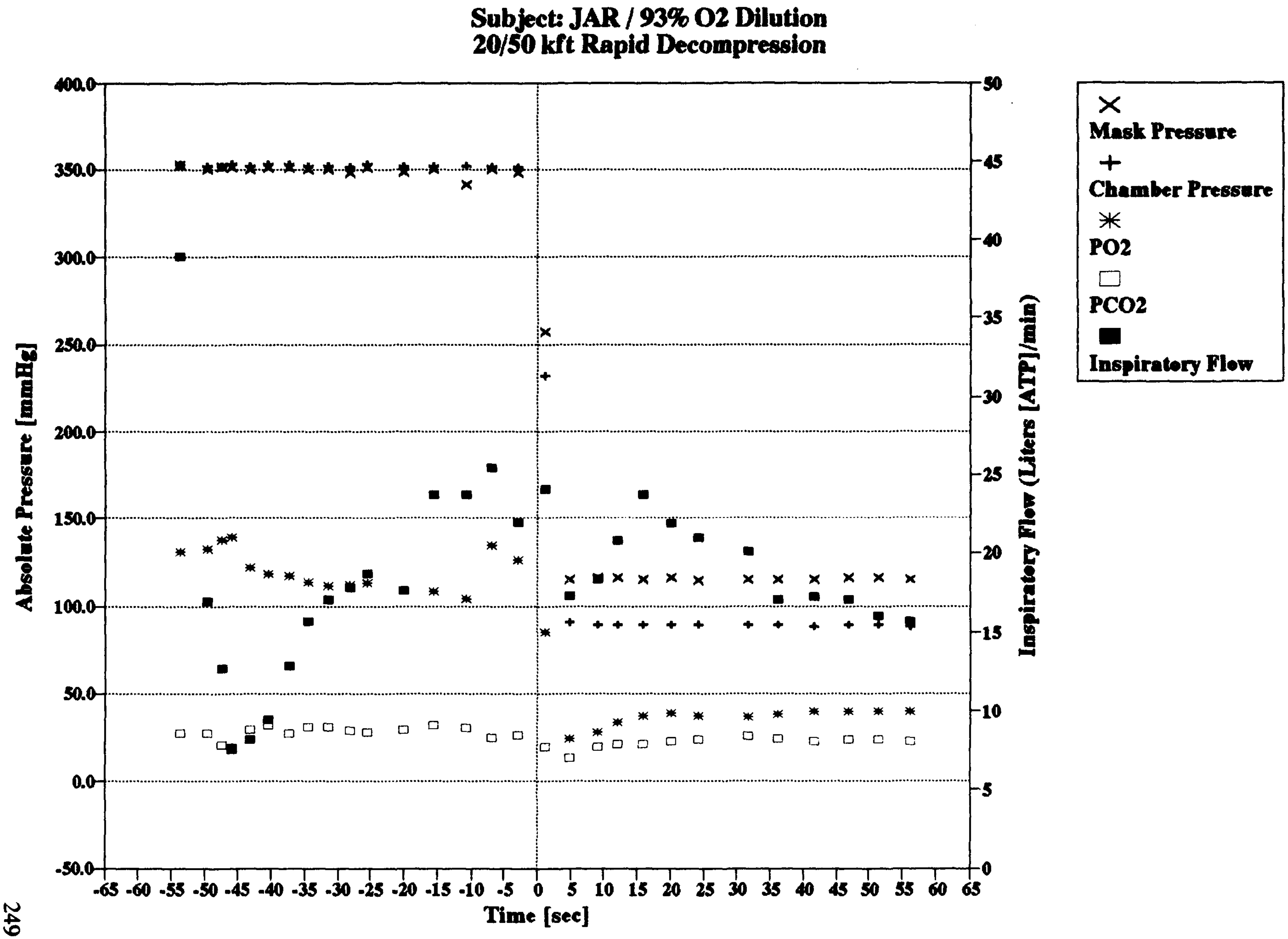

Subject: HOS / 85% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: JAR / 100% O2 Non-Dilution
20/50 kft Rapid Decompression



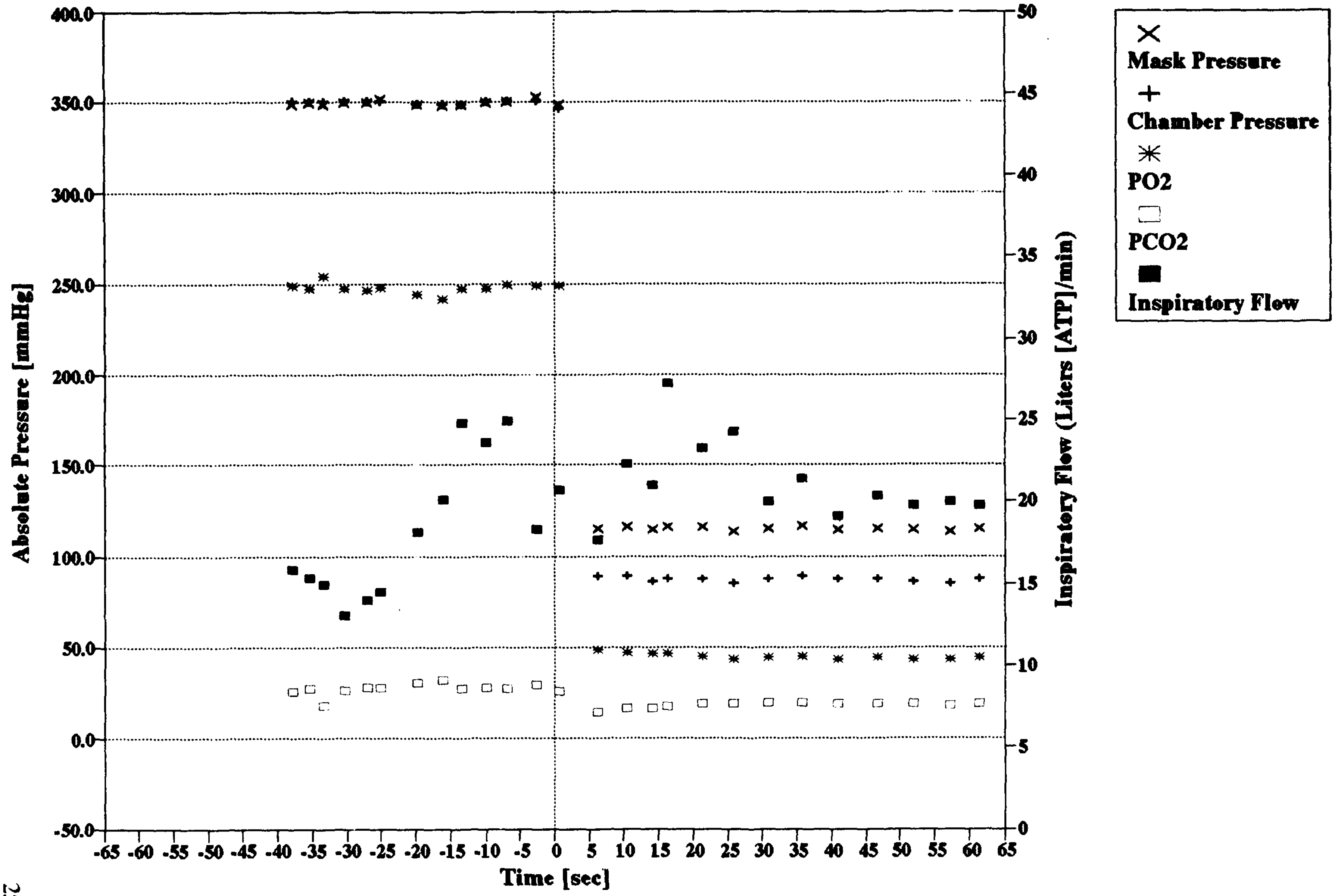


Subject: JAR / 93% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

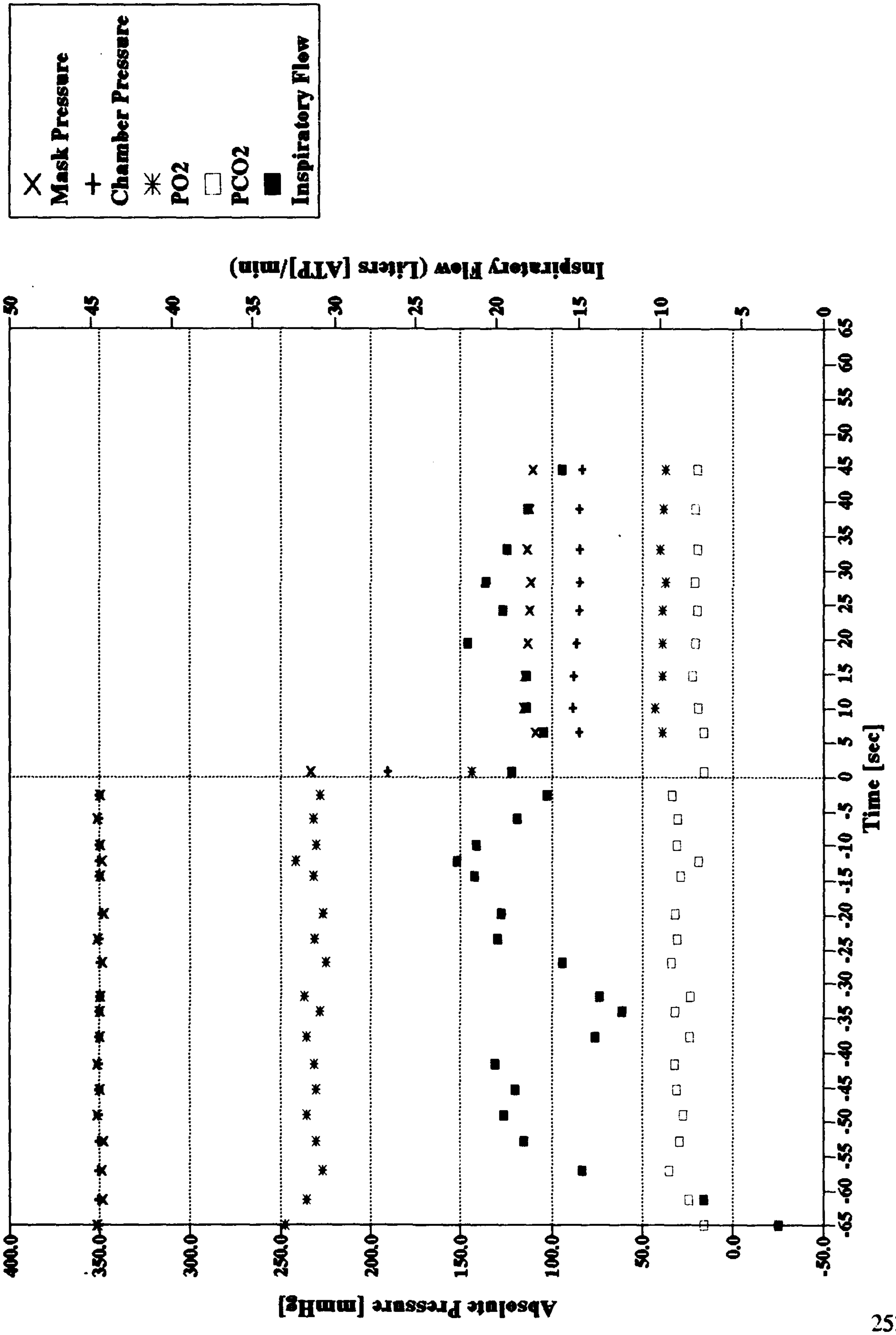


Subject: JAR / 93% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

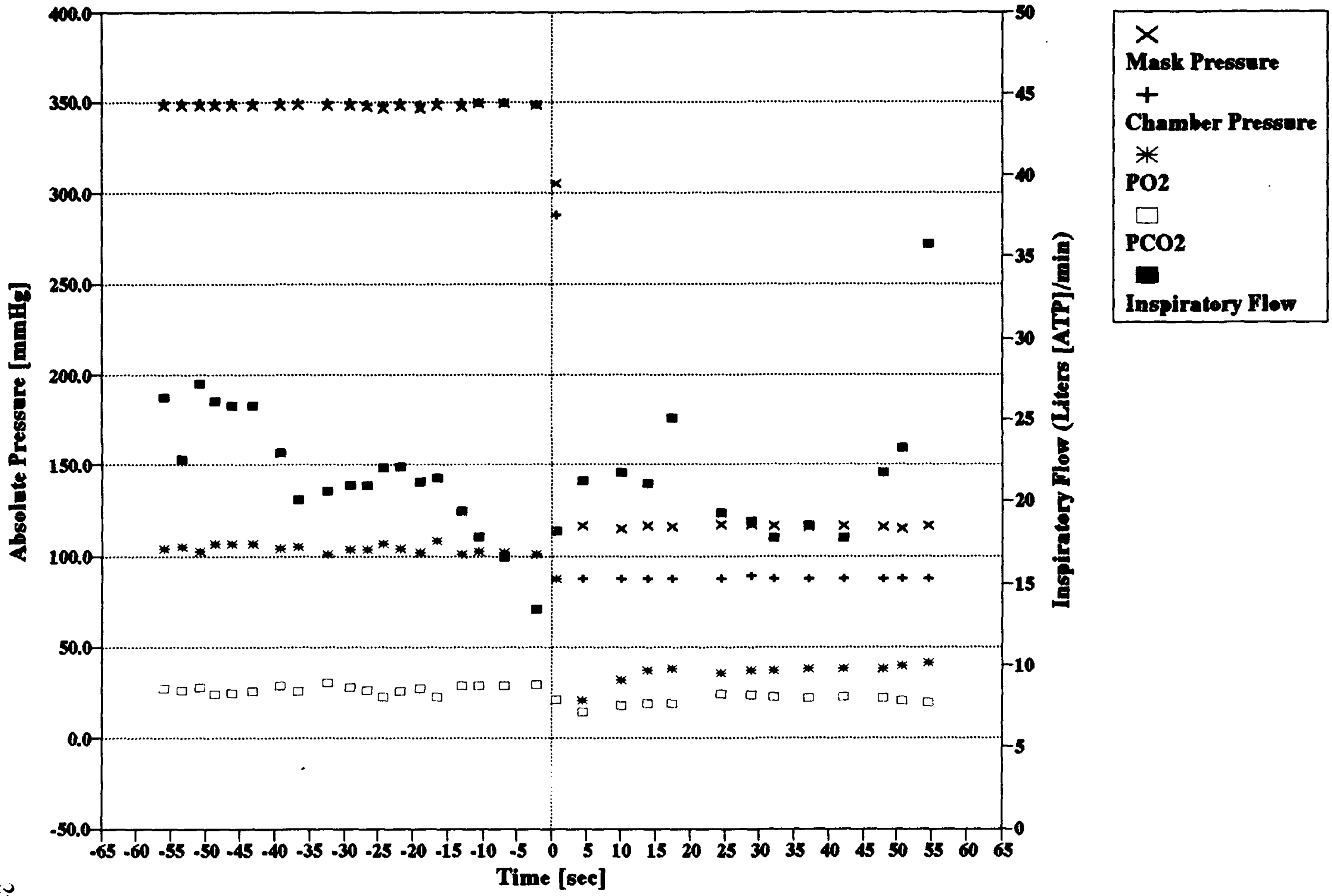
Subject: JAR / 90% O2 Non-Dilution
20/50 kft Rapid Decompression

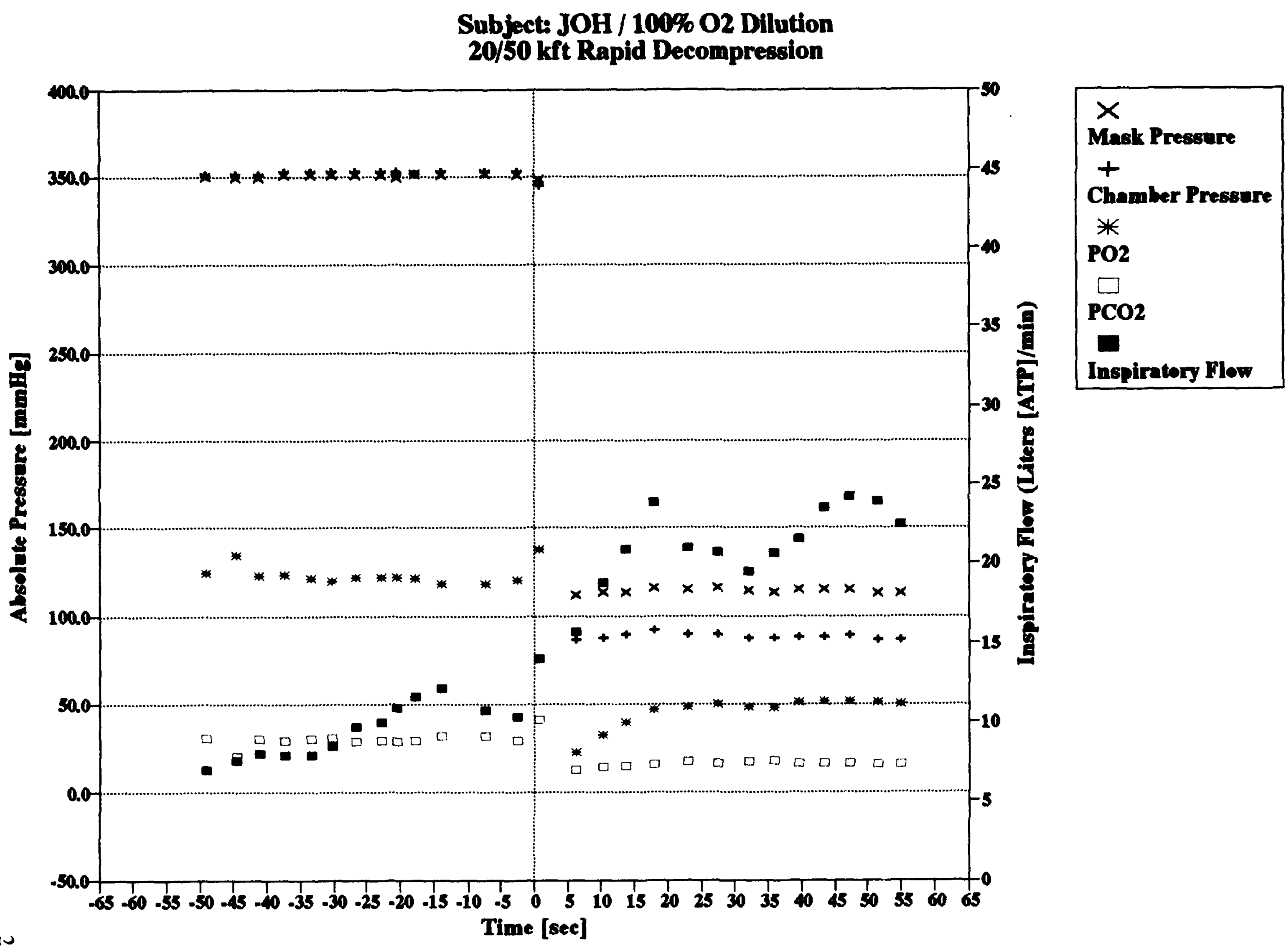


Subject: JAR / 85% O2 Non-Dilution
20/50 kft Rapid Decompression

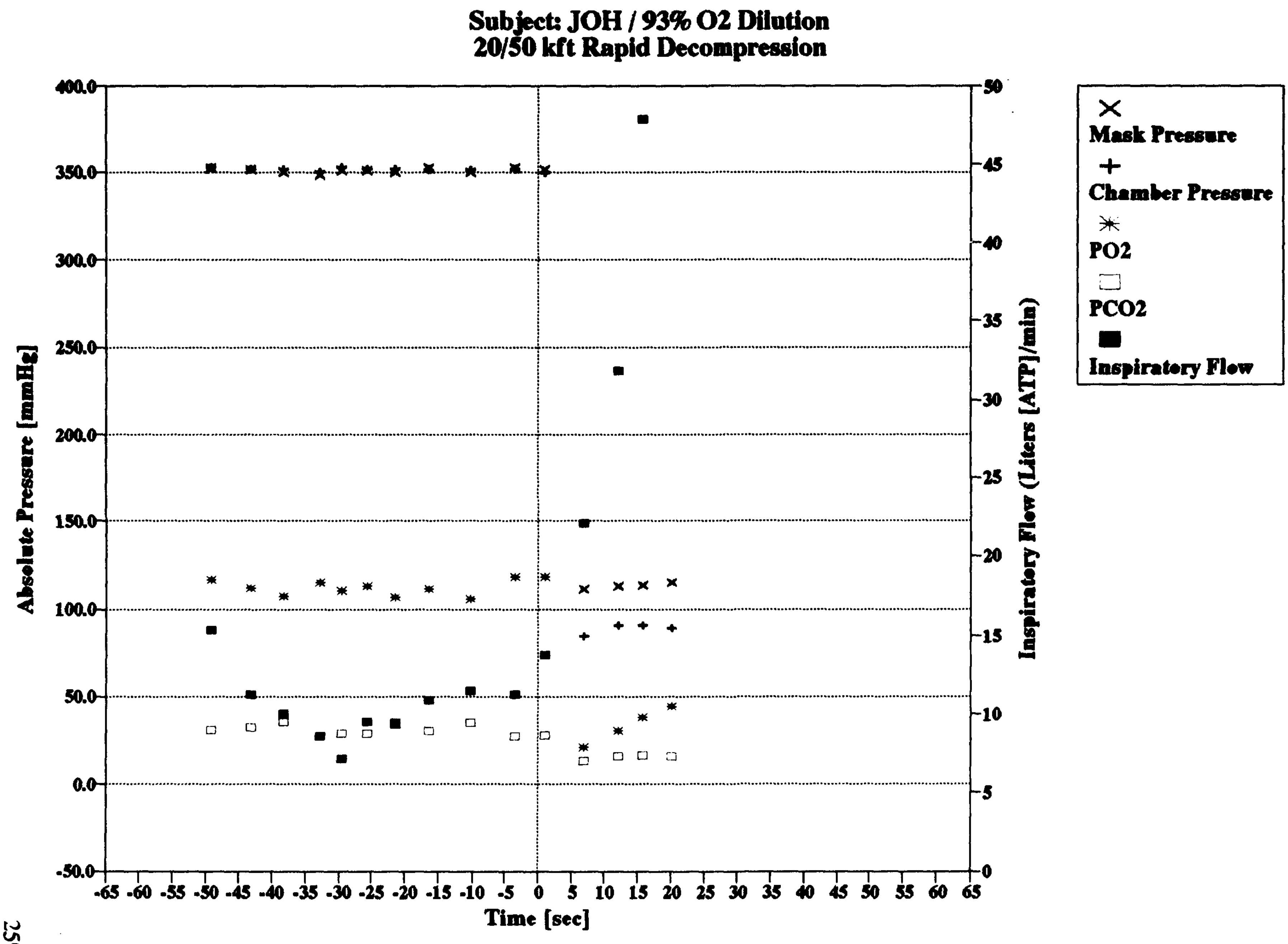


Subject: JAR / 85% O2 Dilution
20/50 kft Rapid Decompression

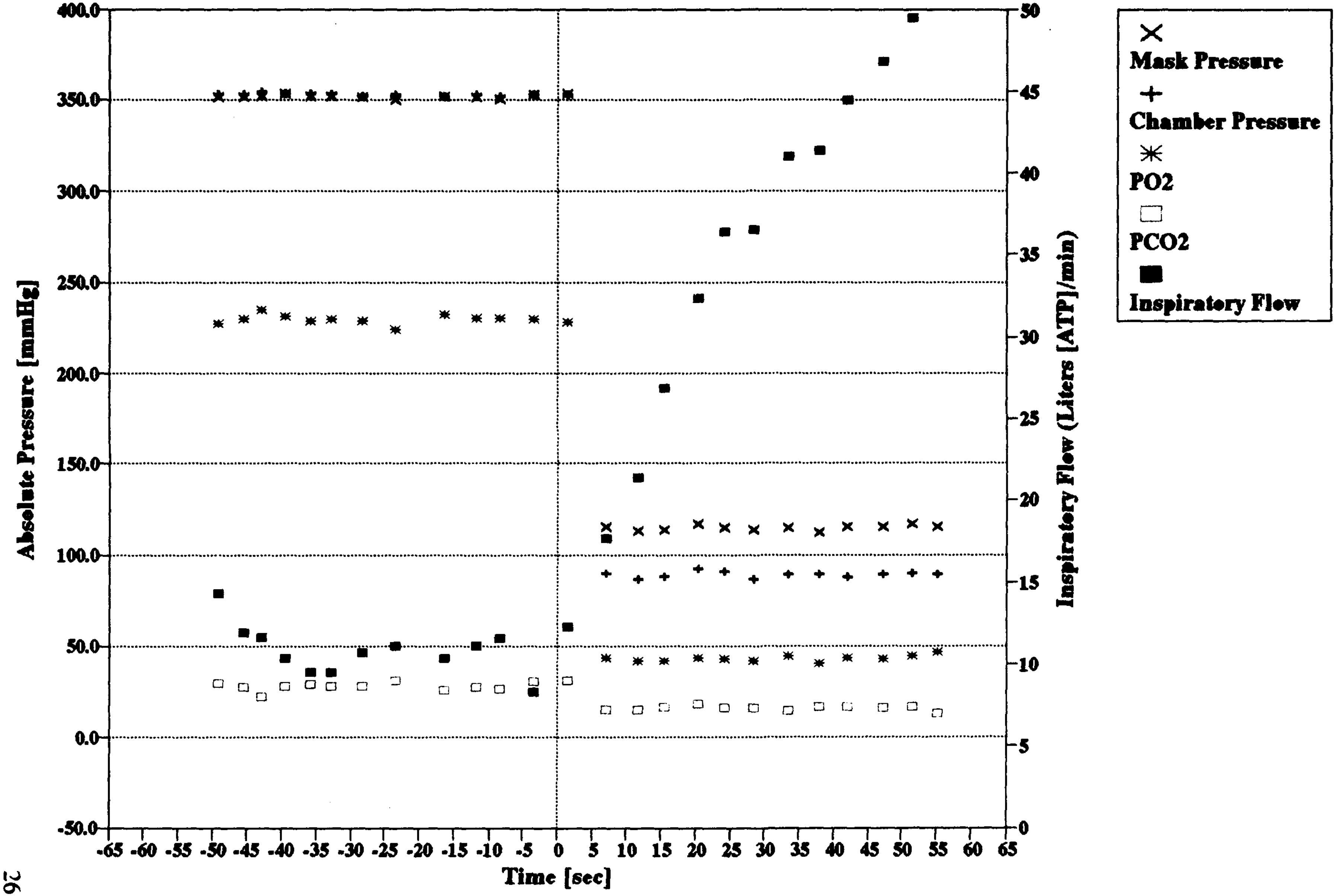




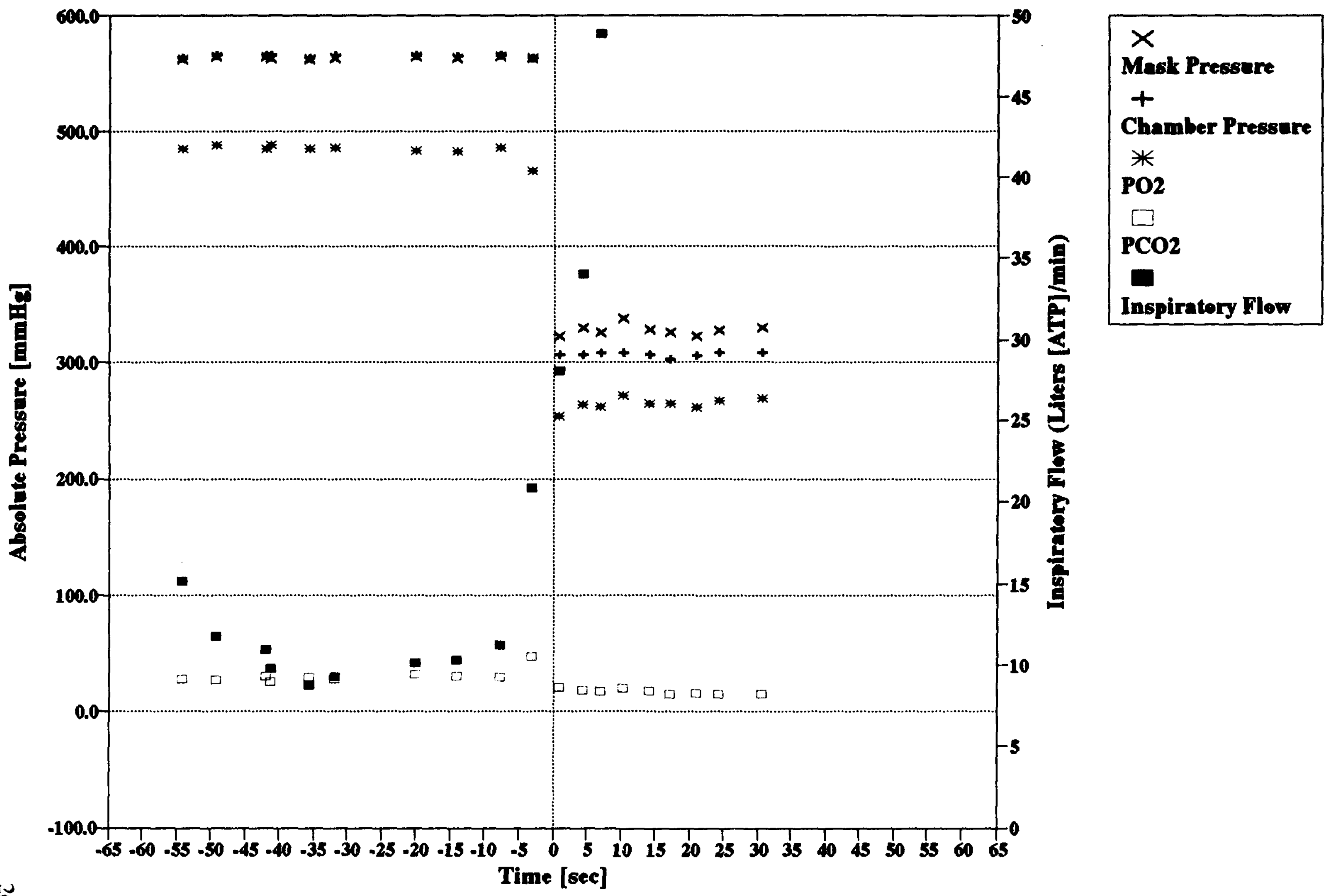
Subject: JOH / 100% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

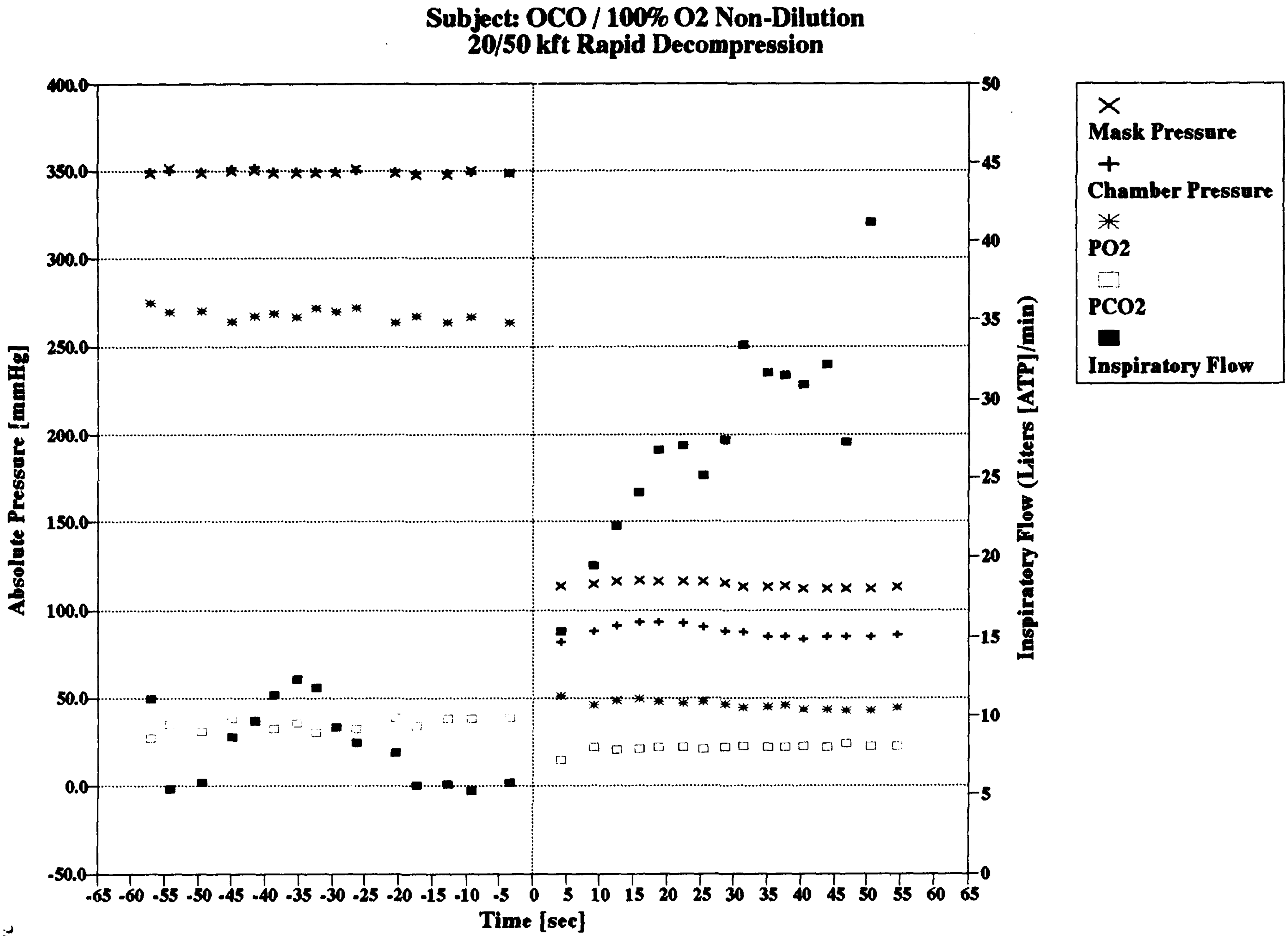

Subject: JOH / 93% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: JOH / 85% O2 Non-Dilution
20/50 kft Rapid Decompression

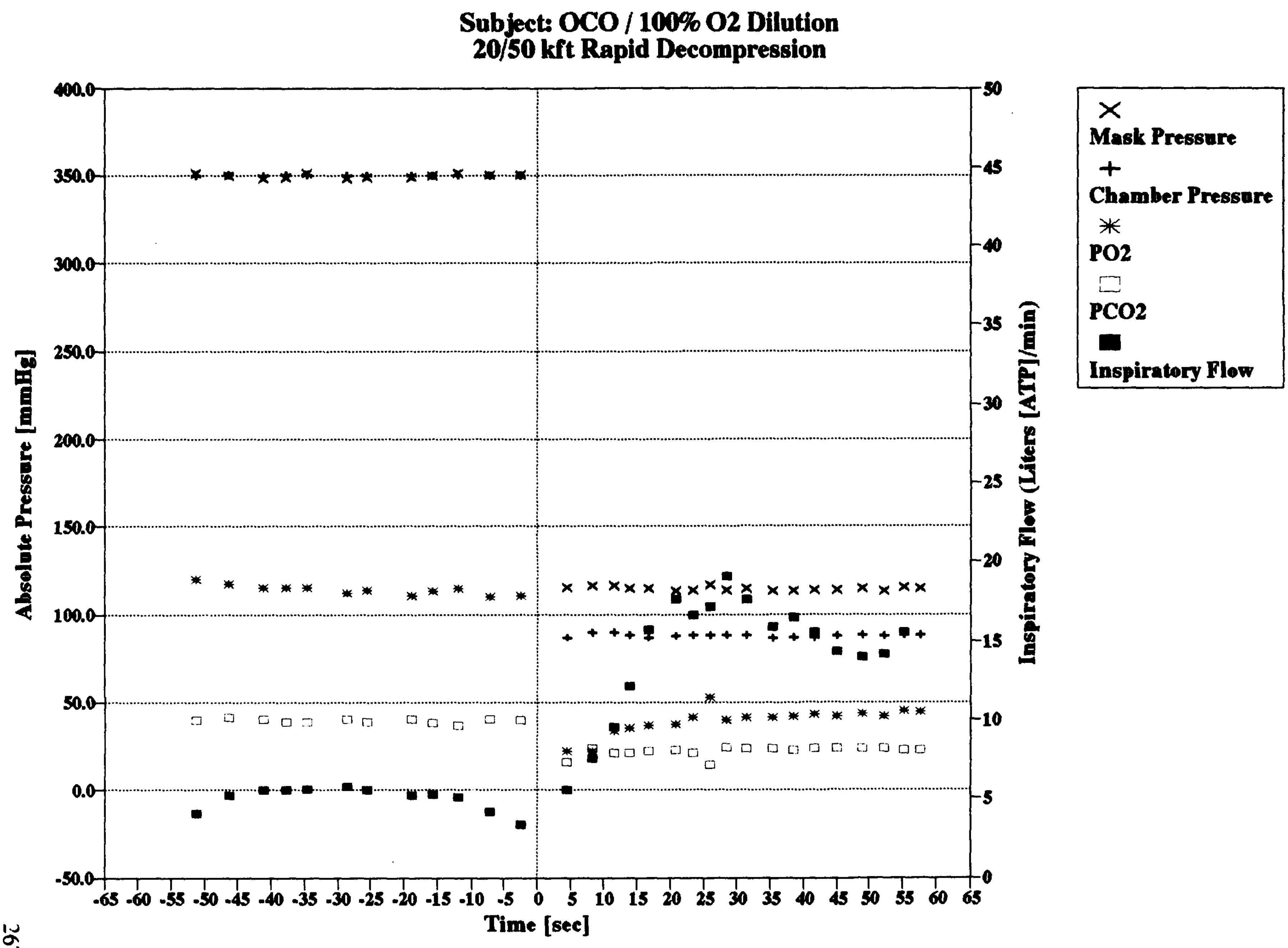


Subject: JOH / 100% O2 Non-Dilution
8/20 kft Rapid Decompression

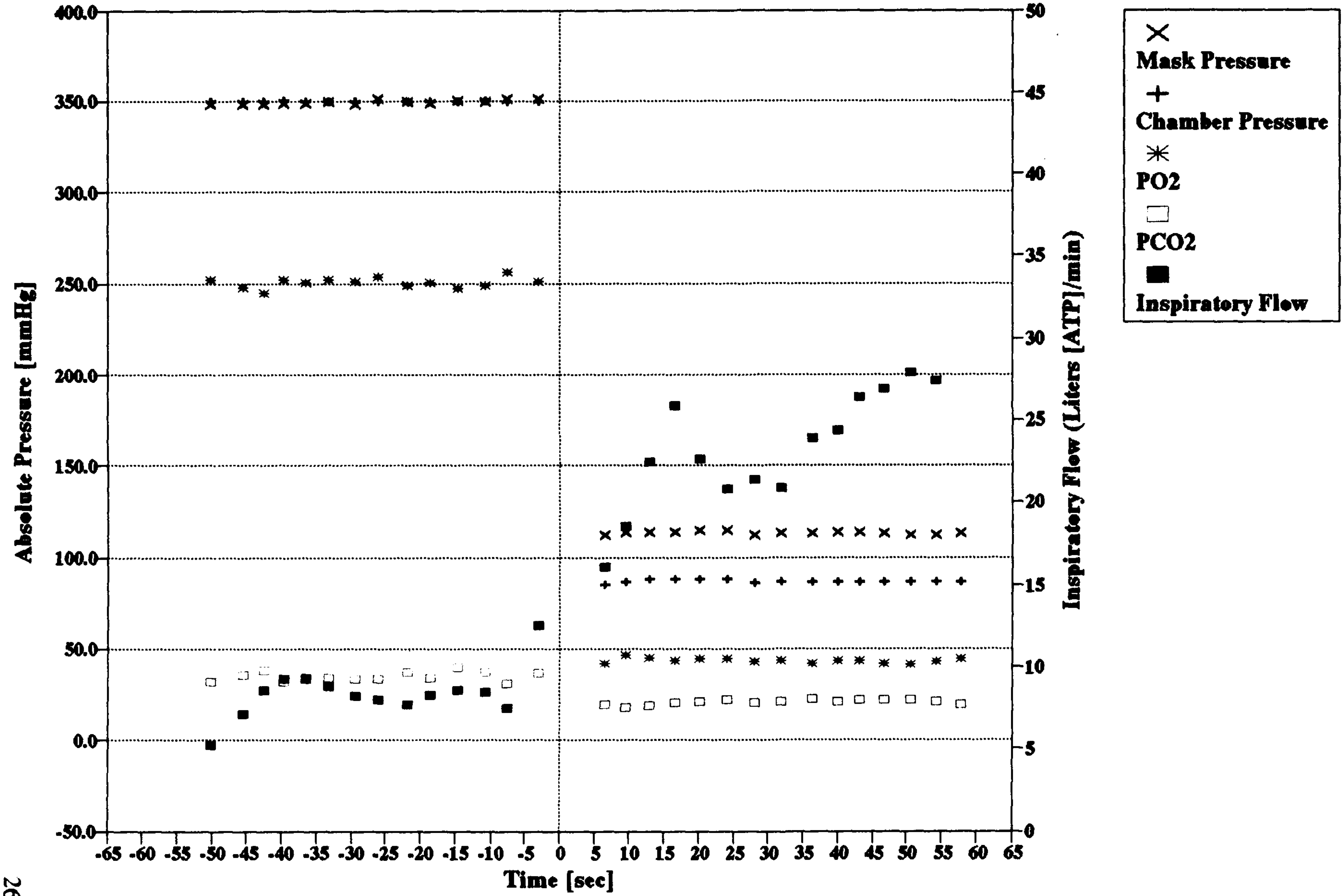




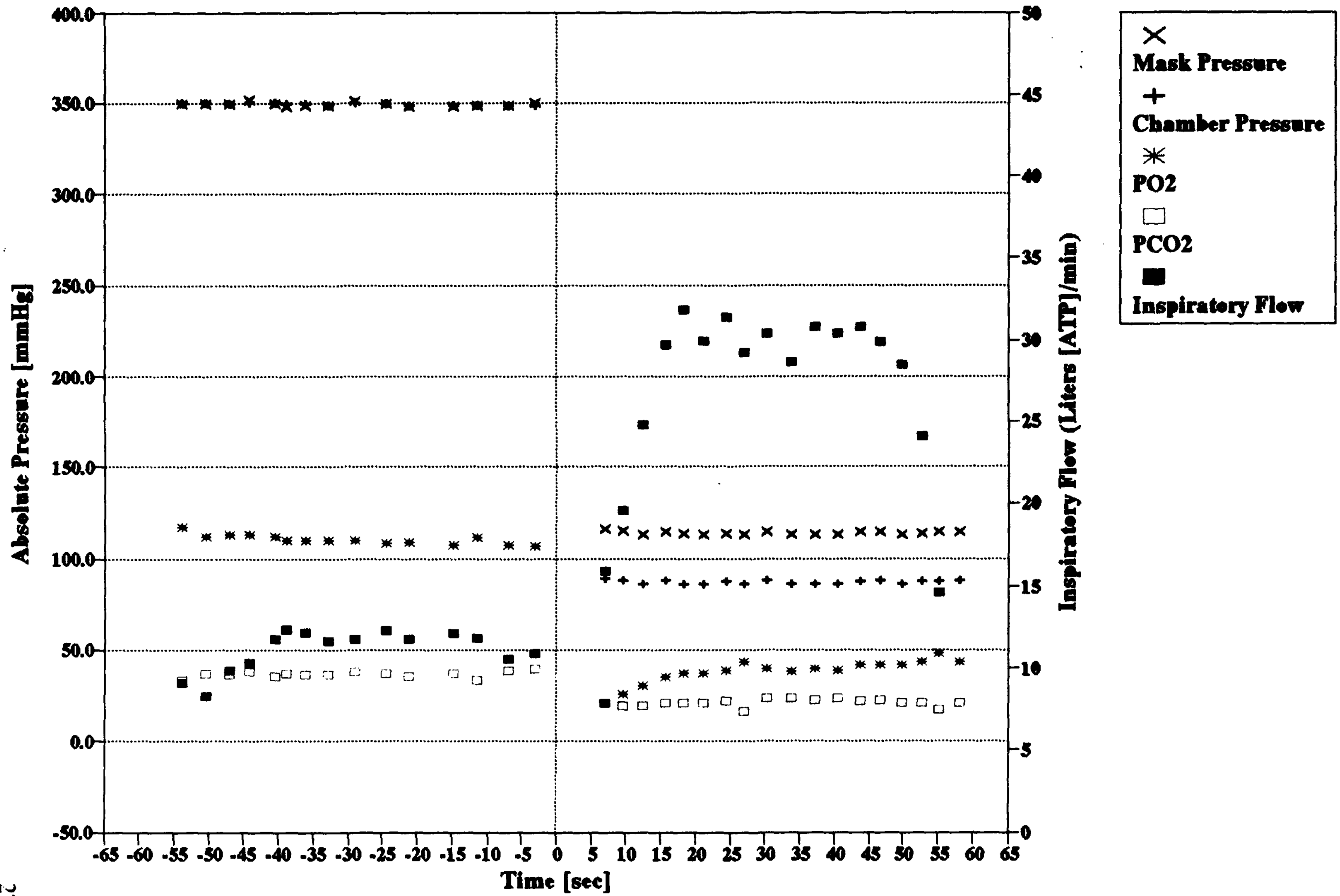
Subject: OCO / 100% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow


Subject: OCO / 100% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

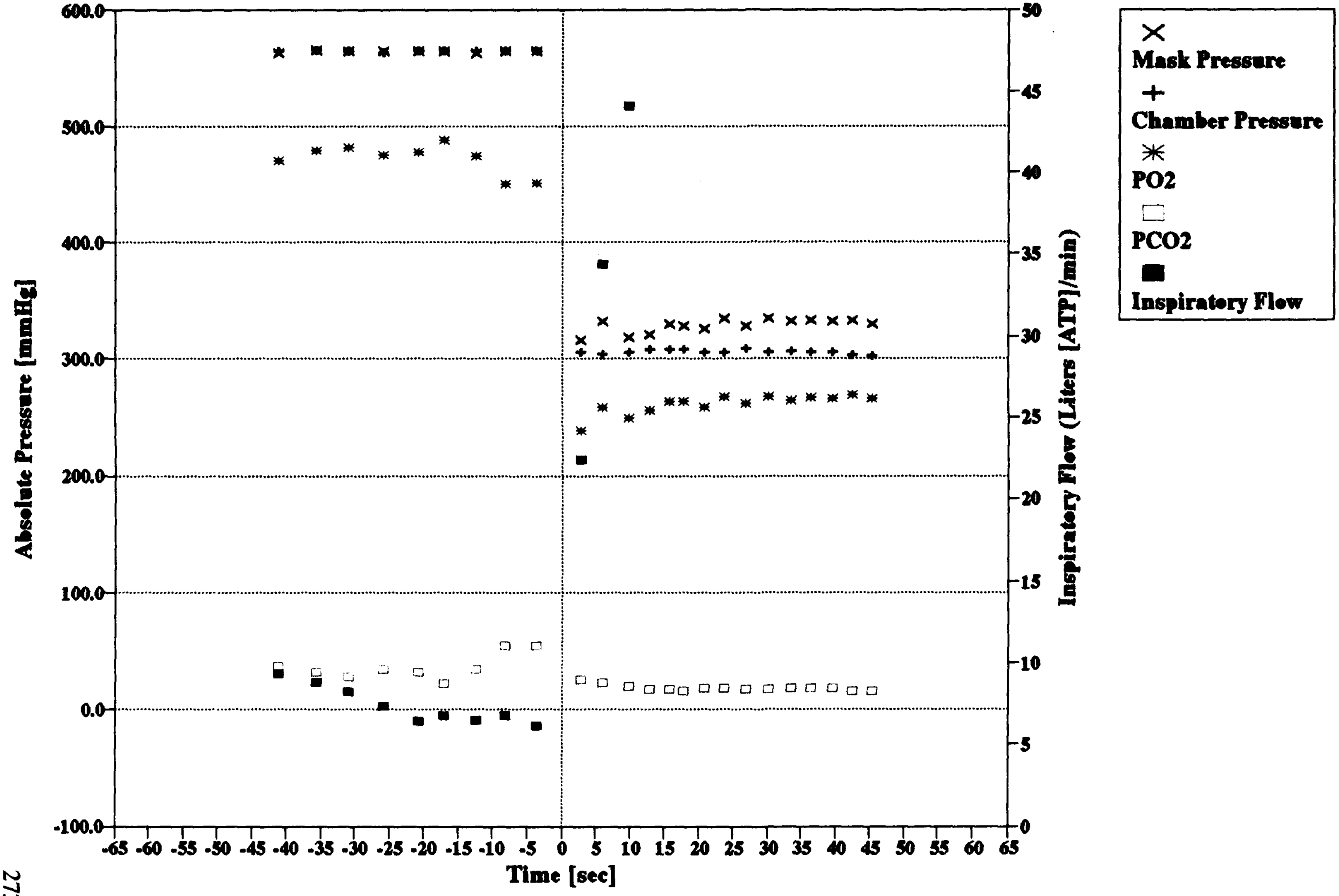
Subject: OCO / 93% O2 Non-Dilution
20/50 kft Rapid Decompression



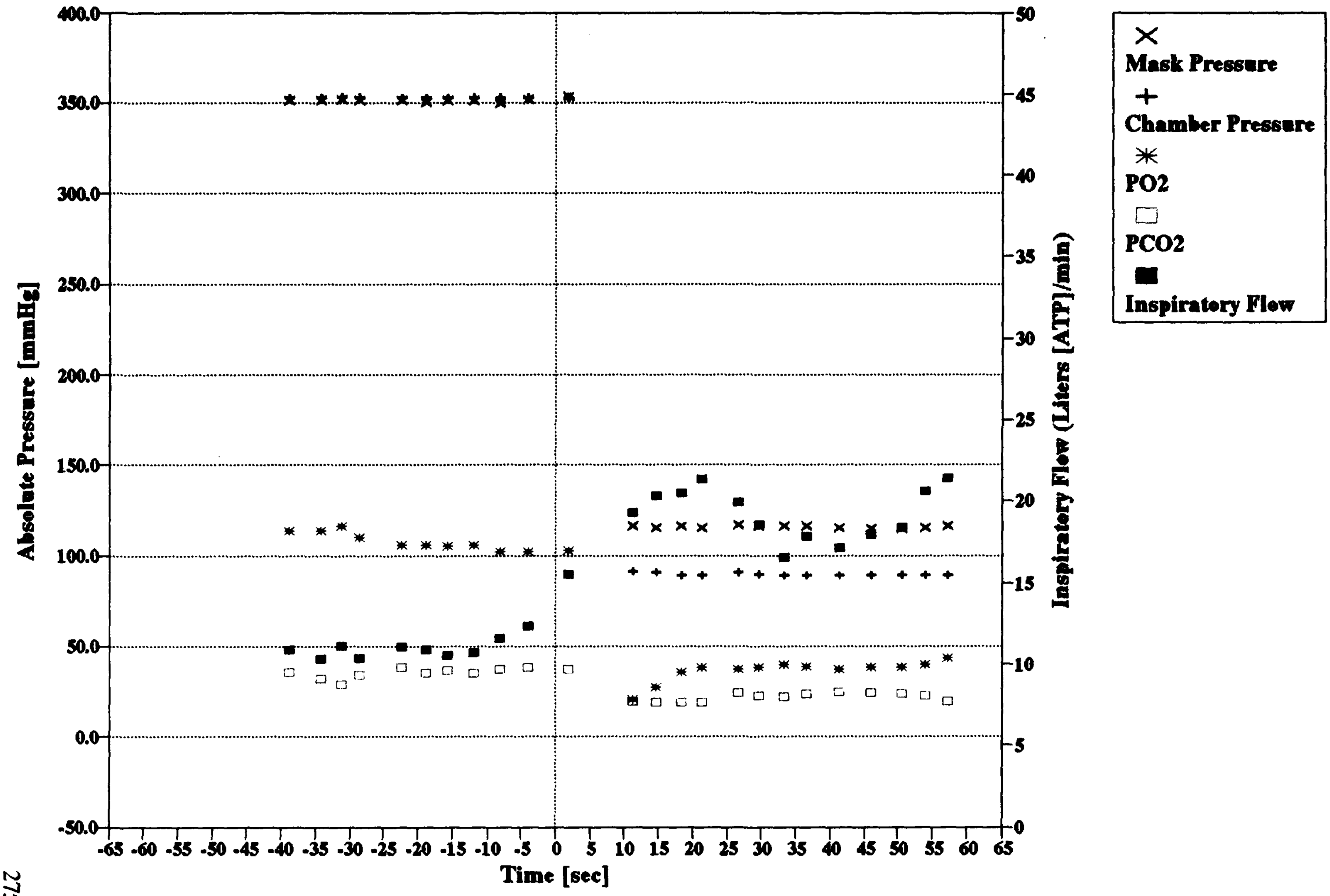
Subject: OCO / 93% O2 Dilution
20/50 kft Rapid Decompression

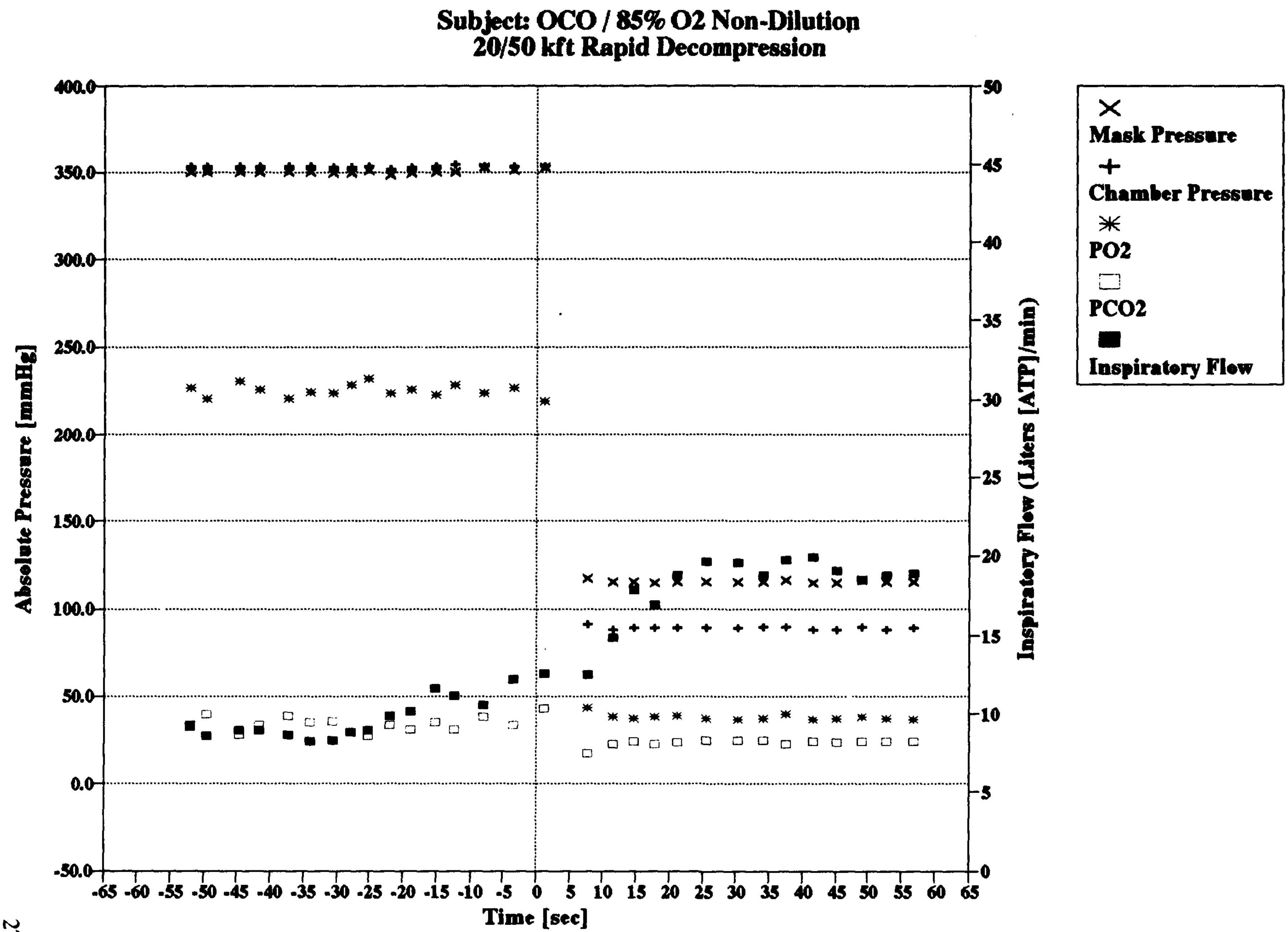


Subject: OCO / 90% O2 Non-Dilution
8/20 kft Rapid Decompression



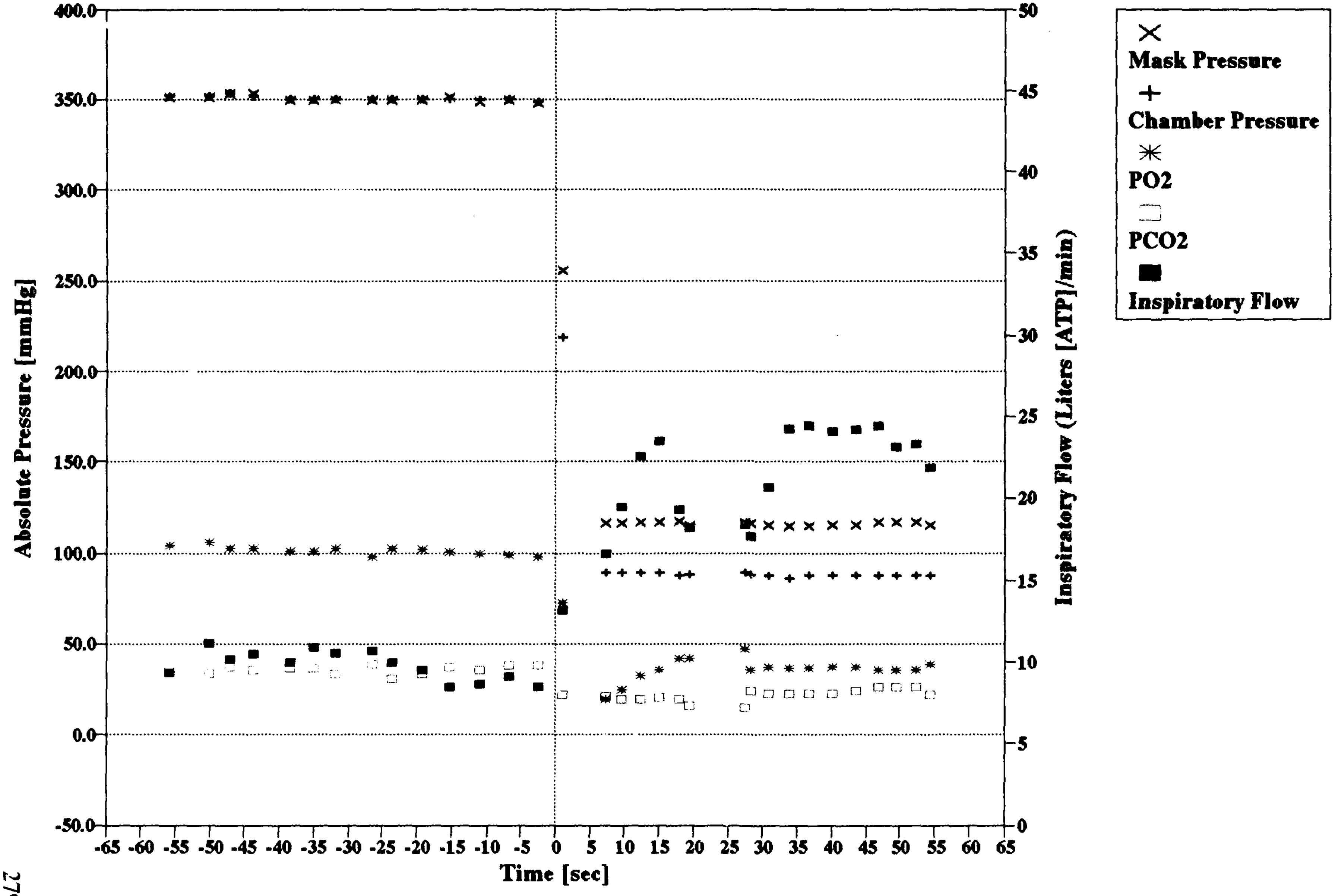
Subject: OCO / 90% O2 Dilution
20/50 kft Rapid Decompression

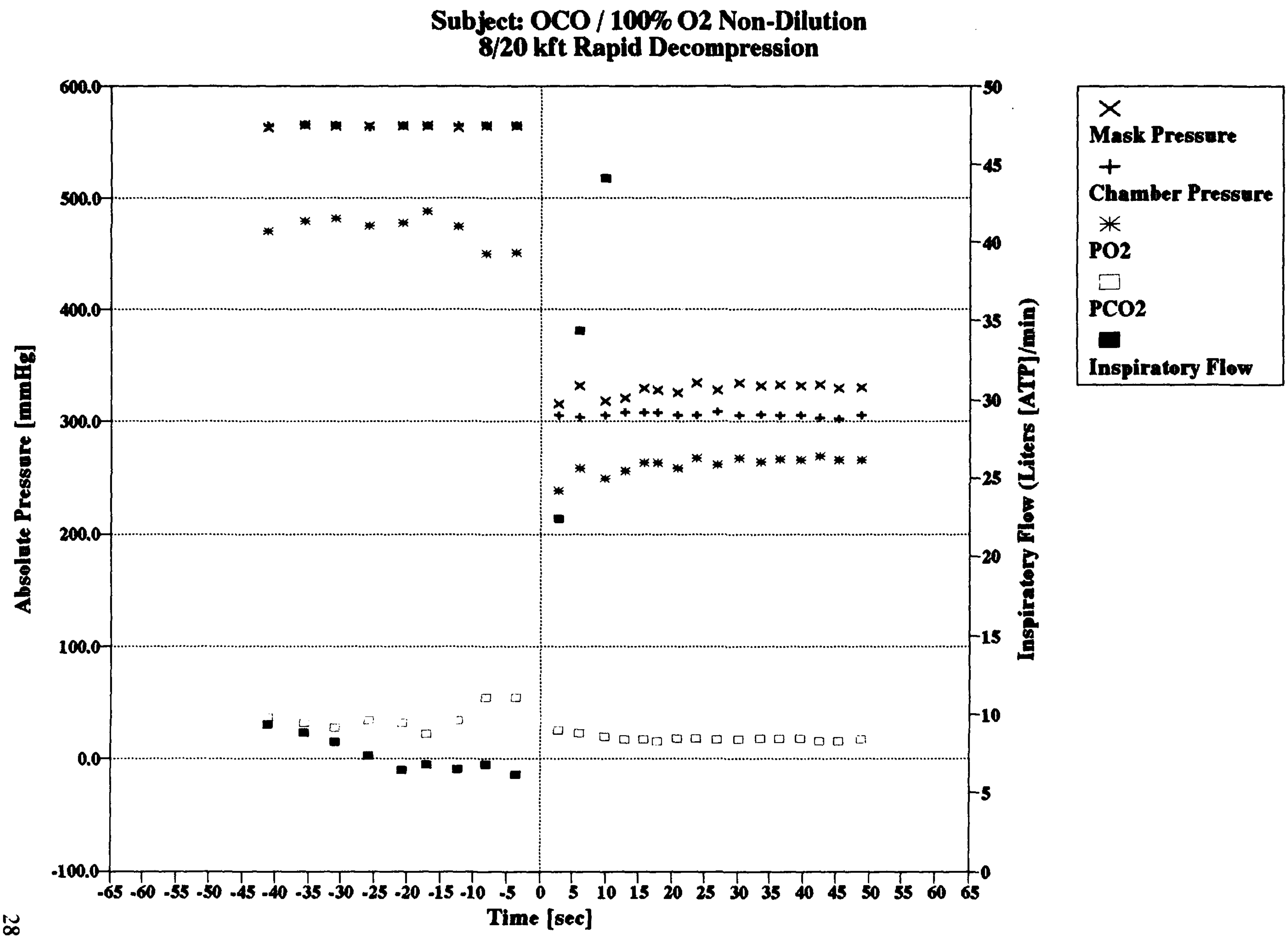




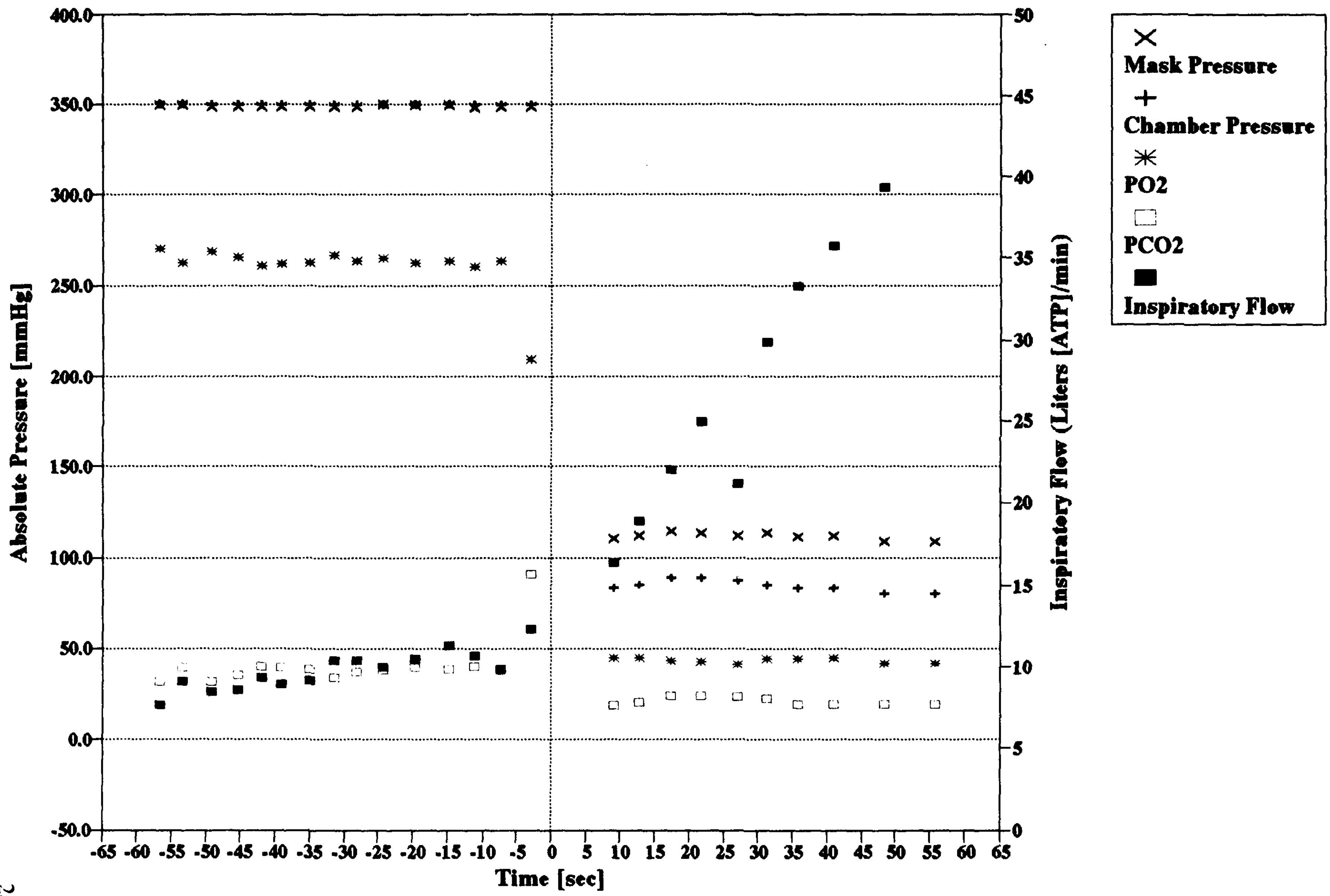
Subject: OCO / 85% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: OCO / 85% O2 Dilution
20/50 kft Rapid Decompression

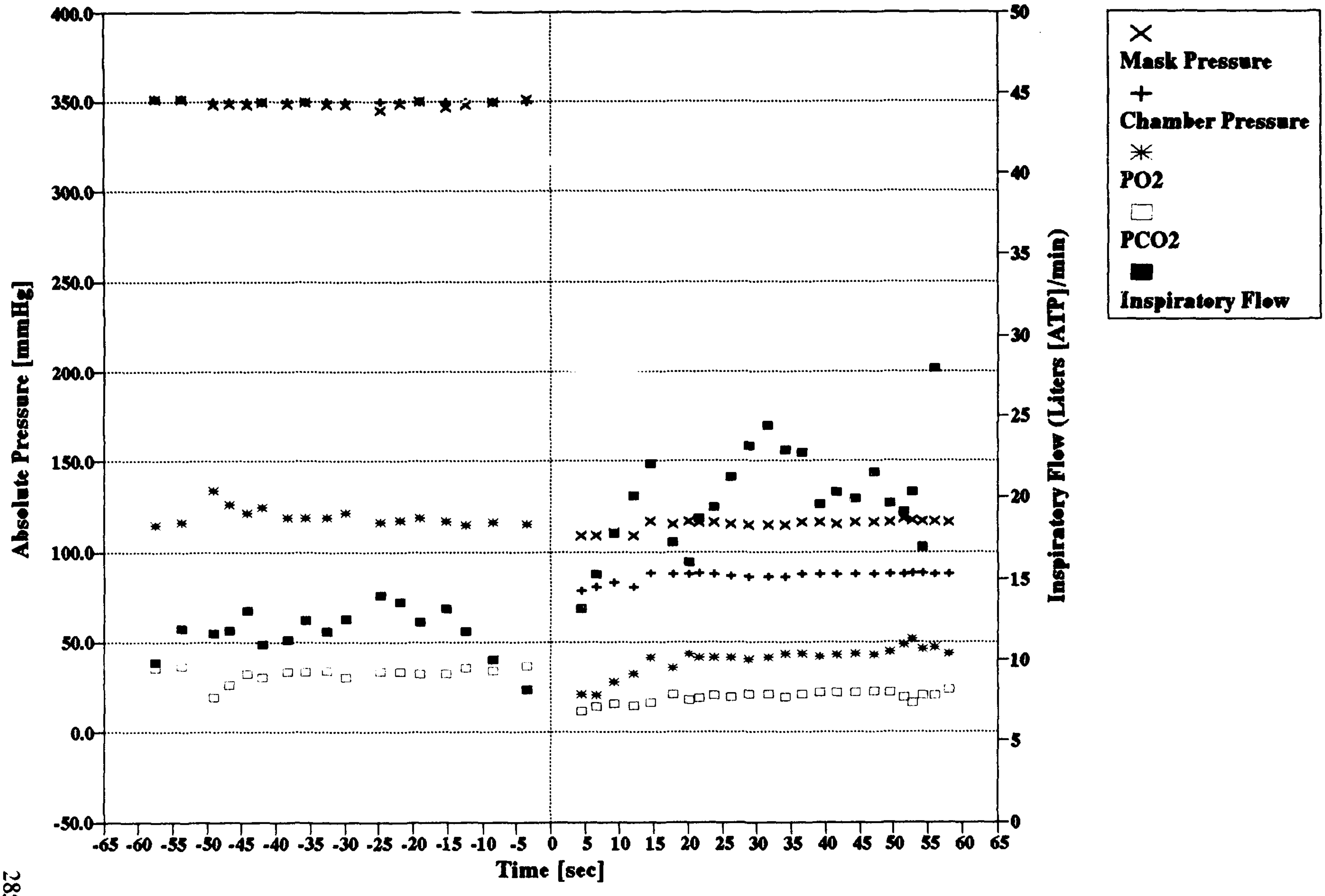



Subject: OCO / 100% O2 Non-Dilution
8/20 kft Rapid Decompression
Absolute Pressure [mmHg]
600.0
500.0
400.0
300.0
200.0
100.0
0.0
-100.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

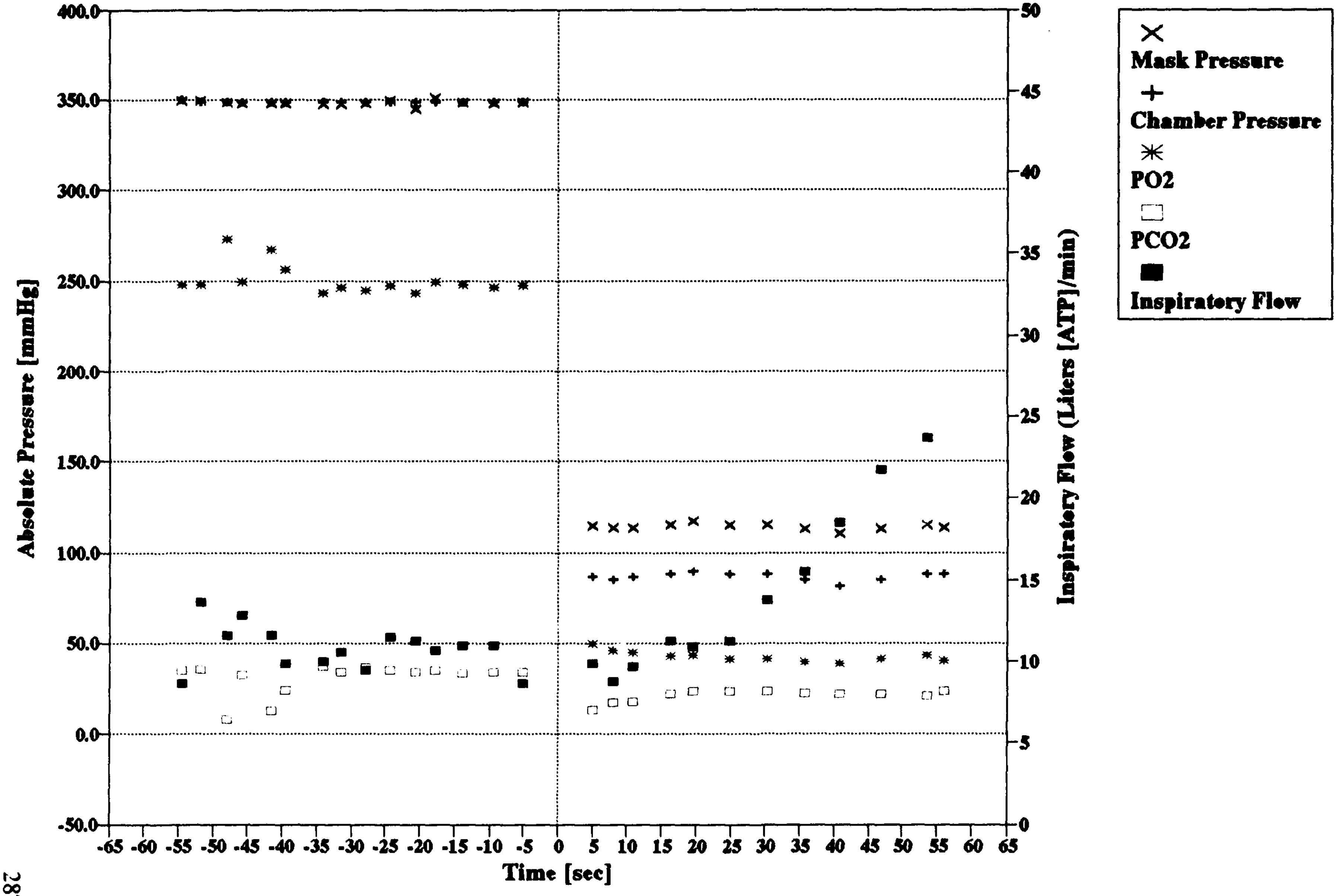
Subject: ORR / 100% O2 Non-Dilution
20/50 kft Rapid Decompression



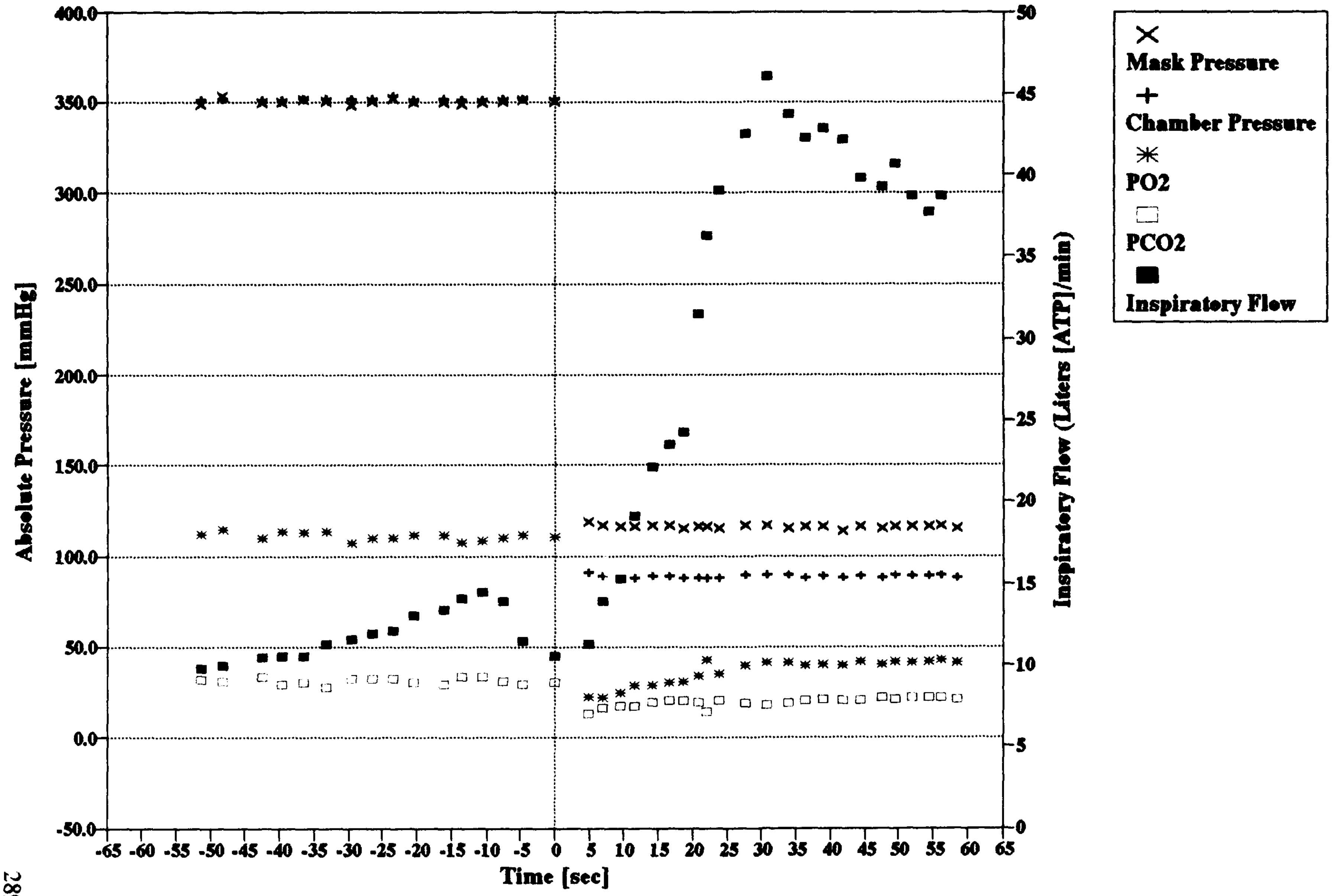
Subject: ORR / 100% O2 Dilution
20/50 kft Rapid Decompression



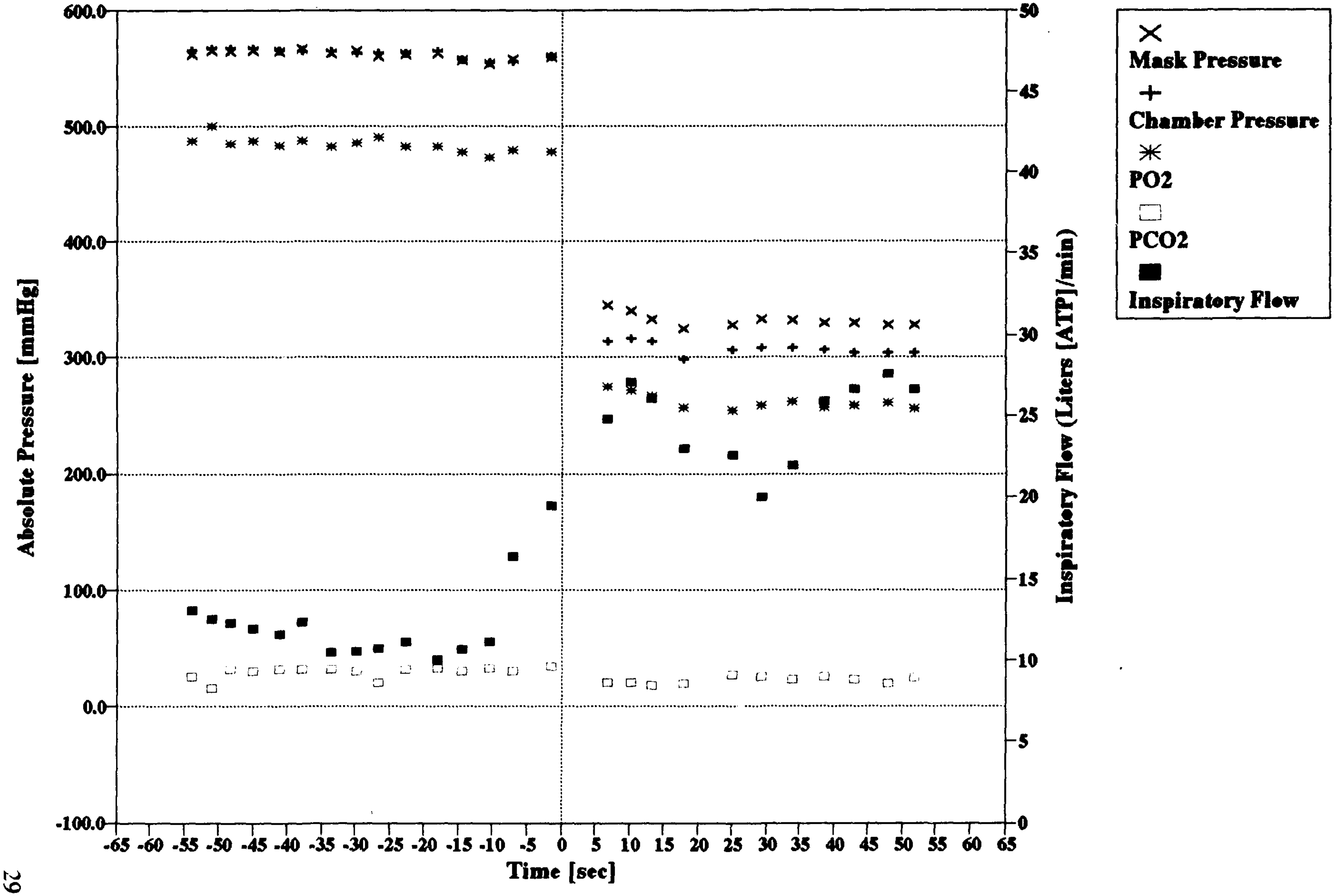
Subject: ORR / 93% O2 Non-Dilution
20/50 kft Rapid Decompression

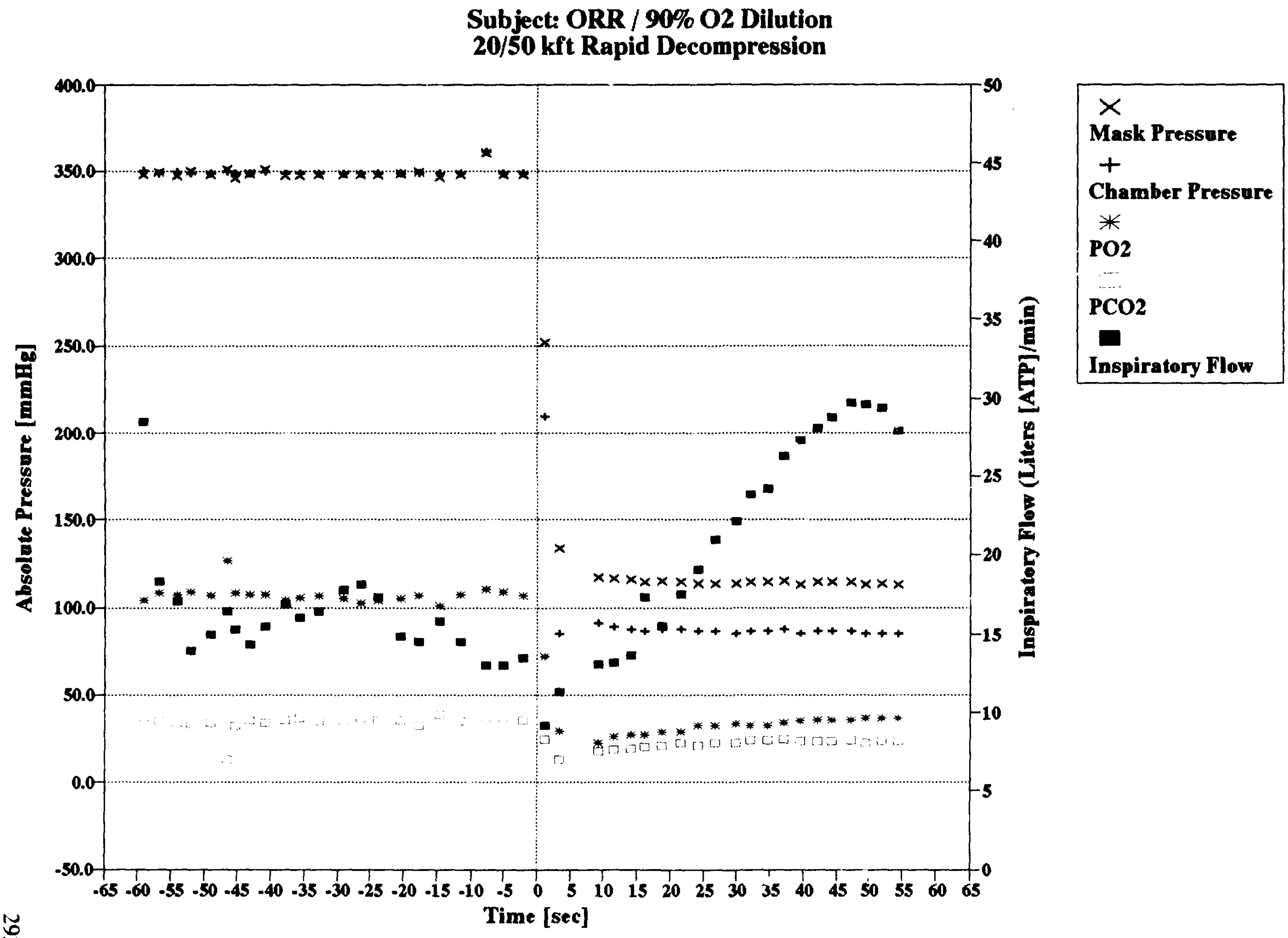


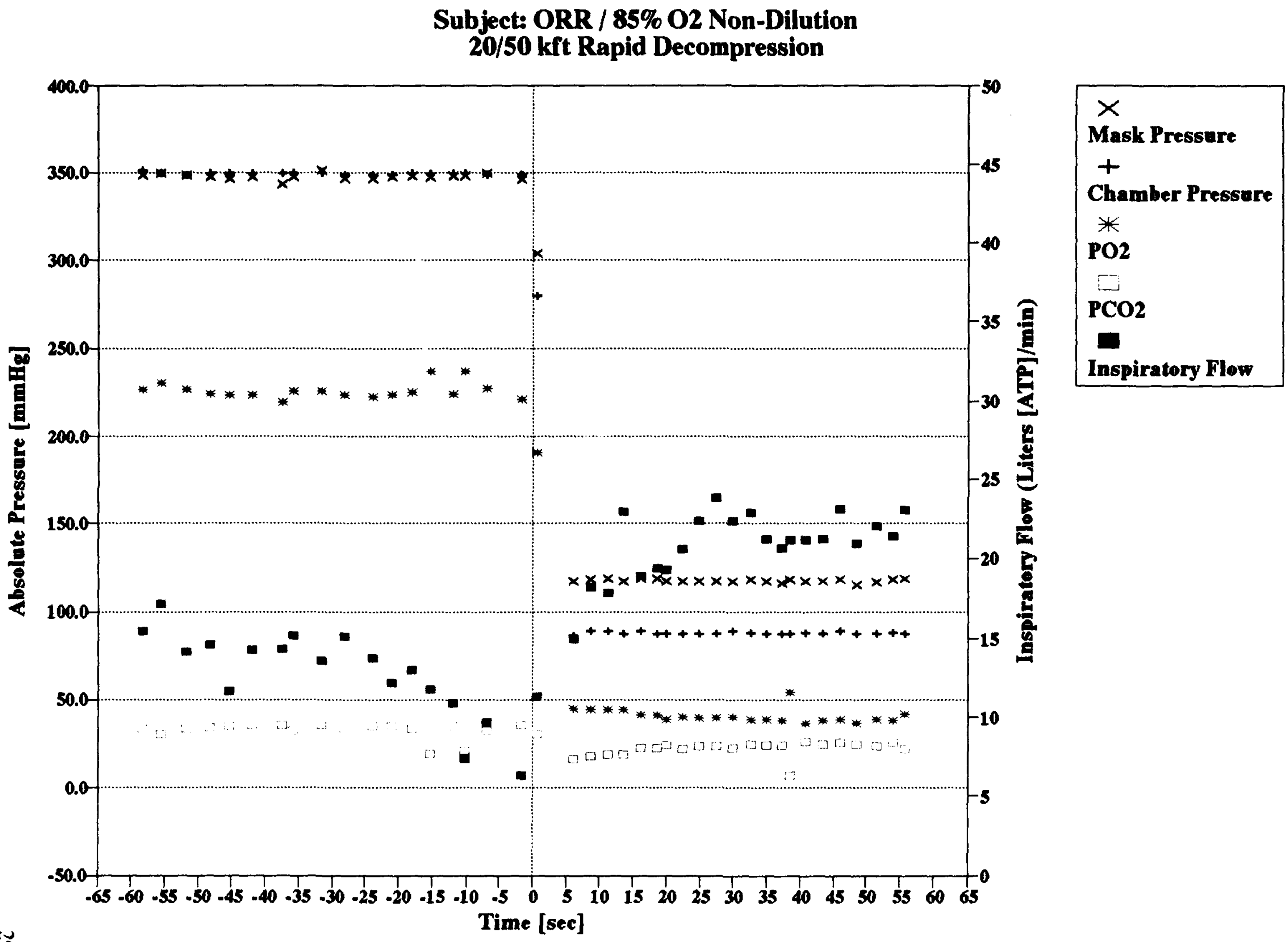
Subject: ORR / 93% O2 Dilution
20/50 kft Rapid Decompression

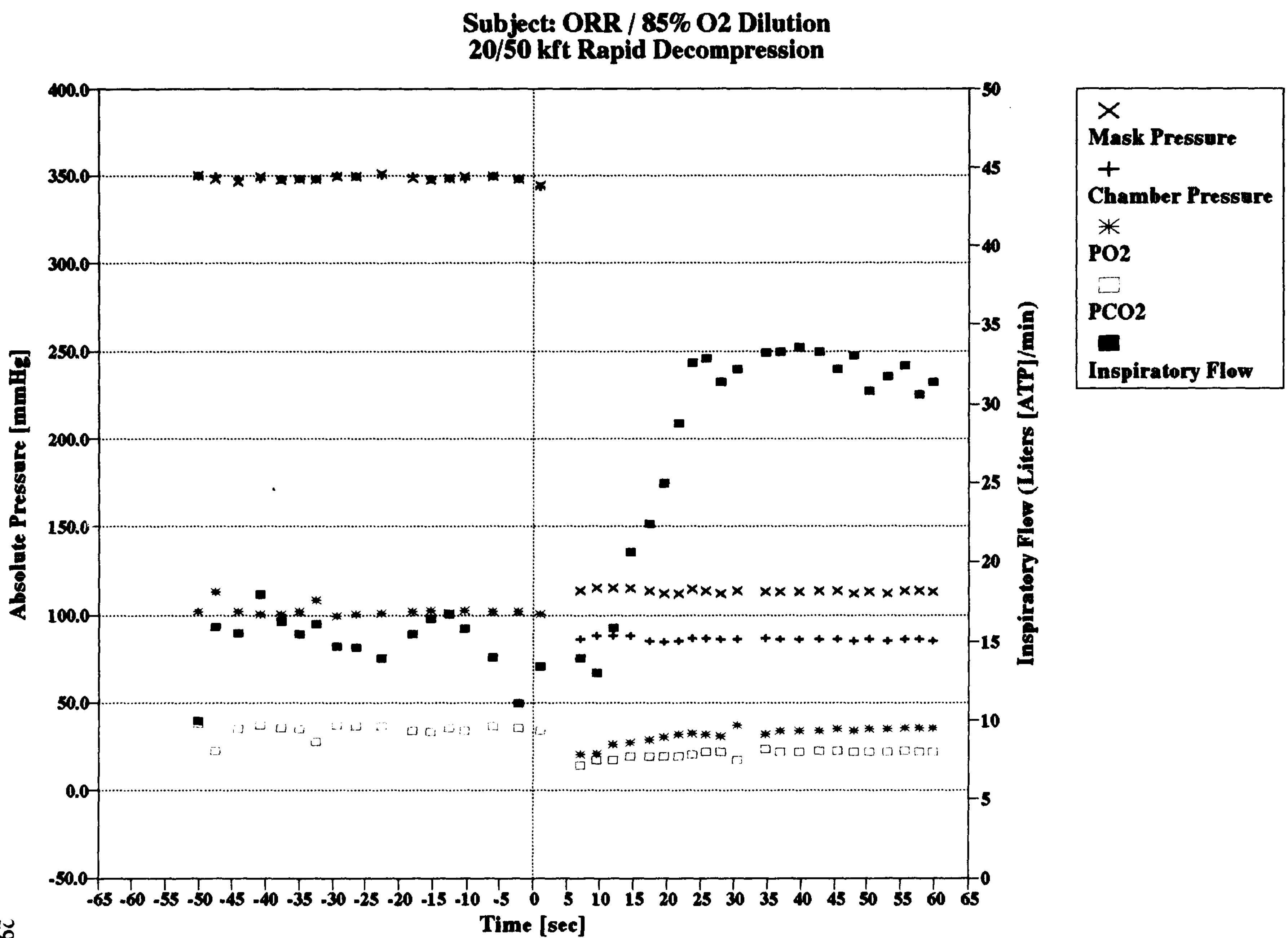


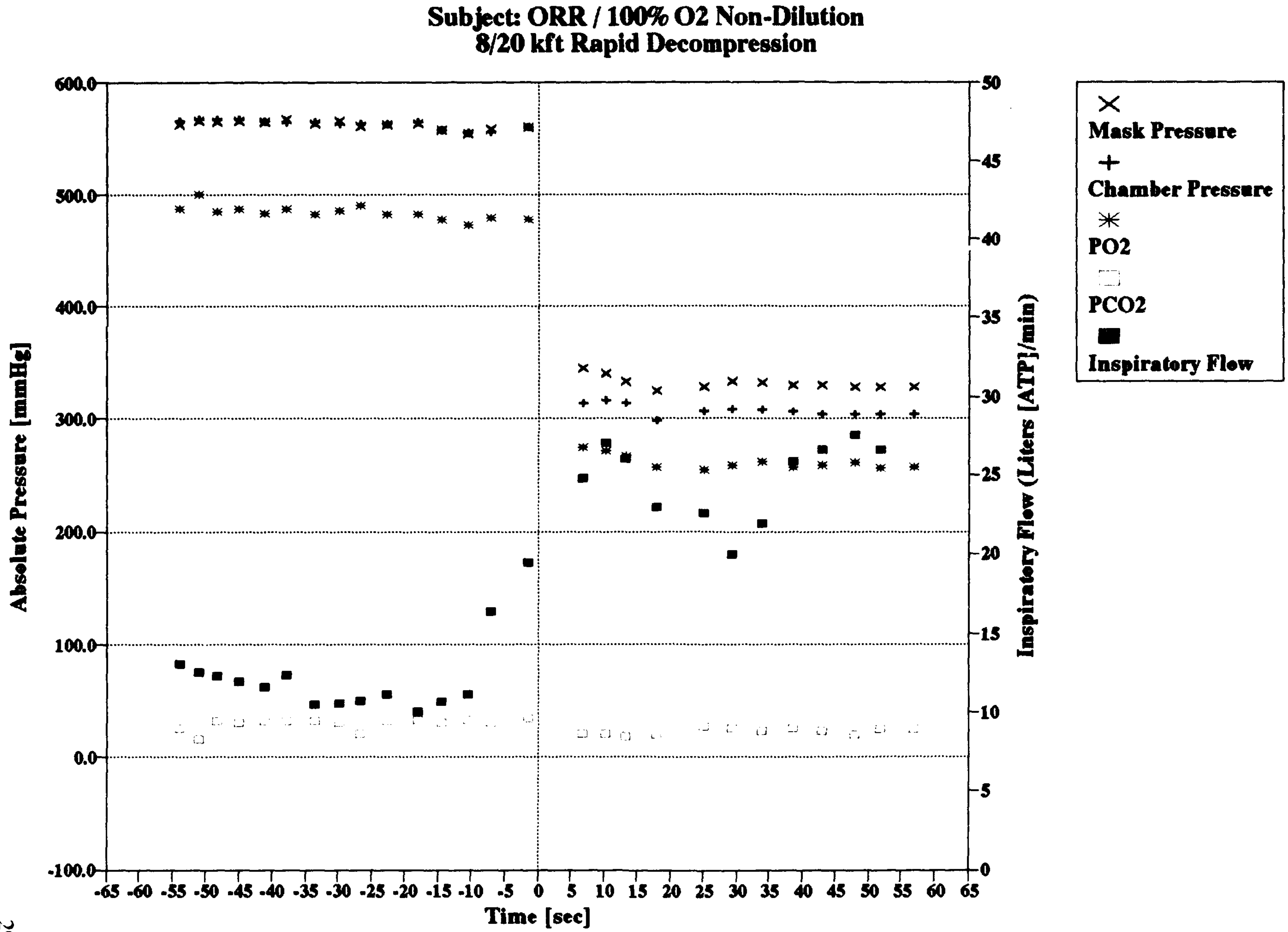
Subject: ORR / 90% O2 Non-Dilution
8/20 kft Rapid Decompression



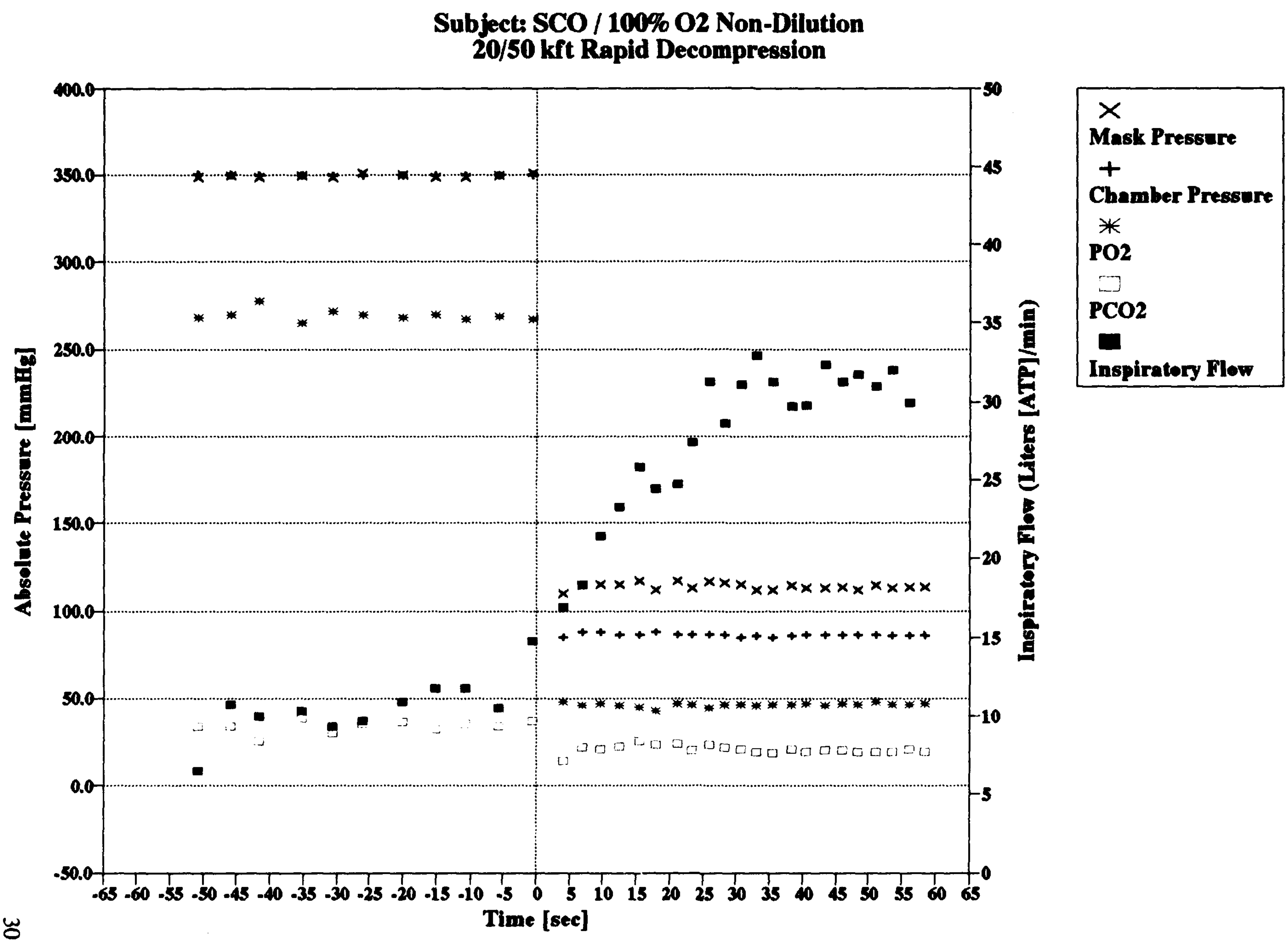

Subject: ORR / 90% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow


Subject: ORR / 85% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

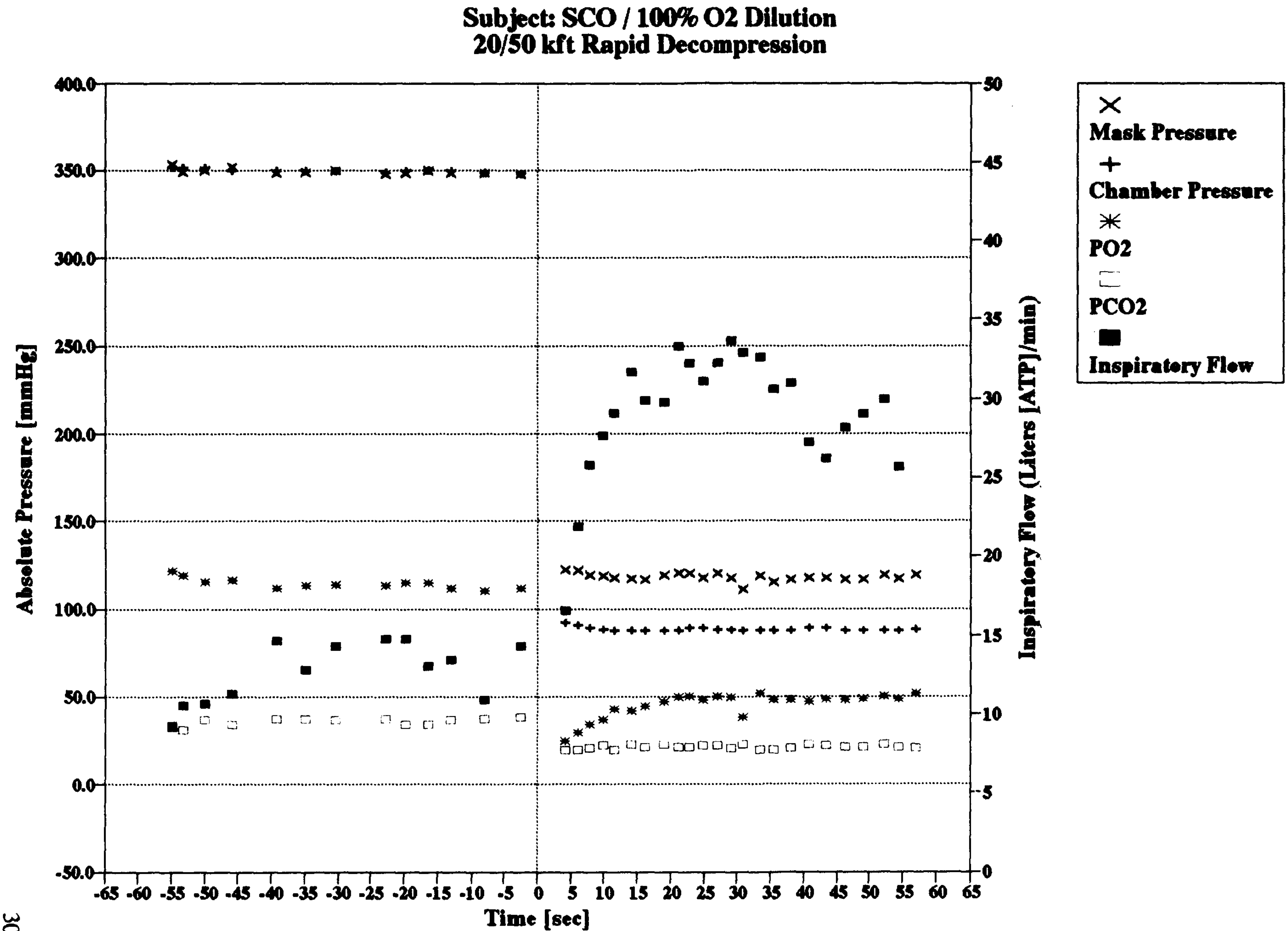

Subject: ORR / 85% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

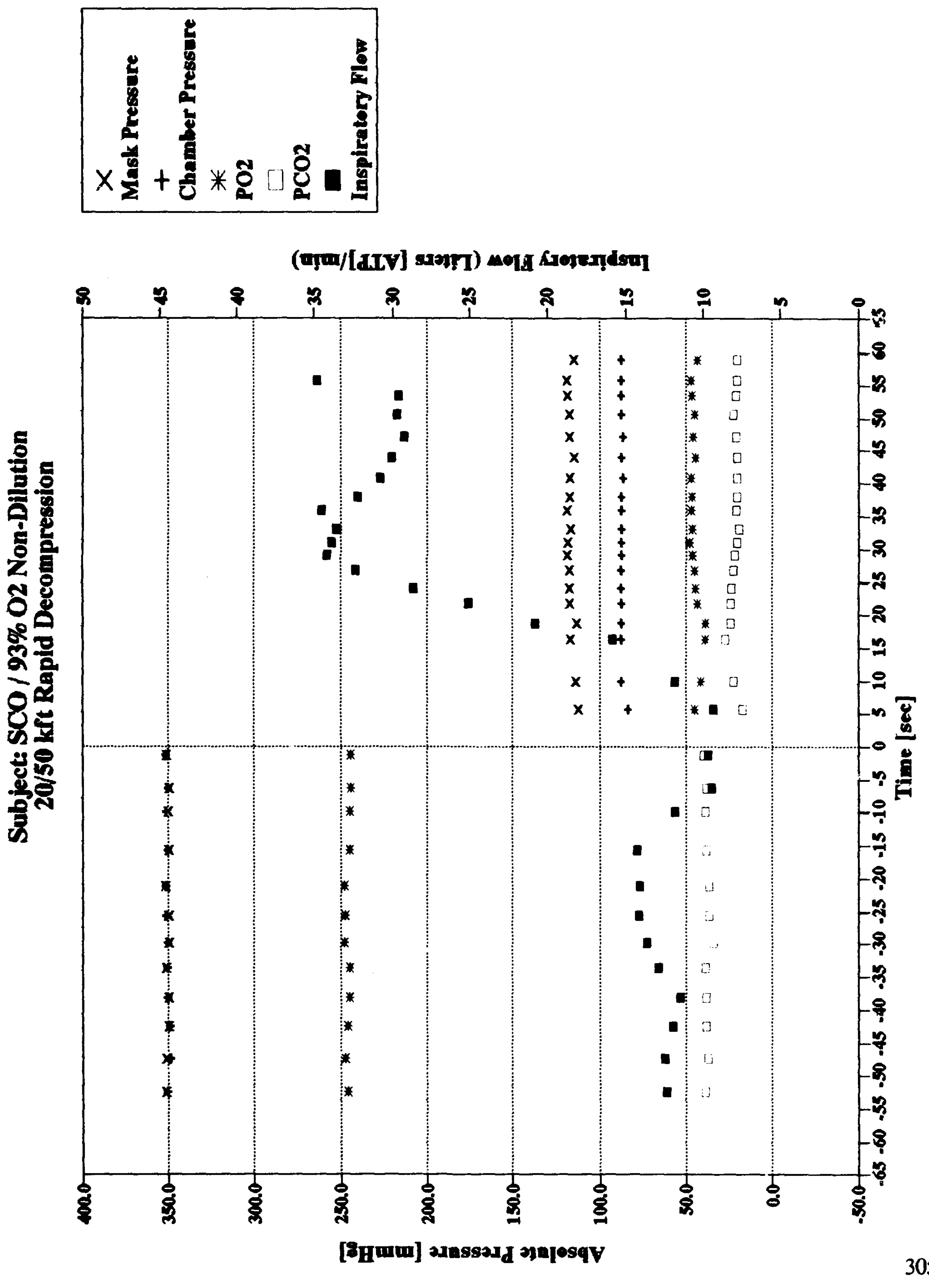


Subject: ORR / 100% O2 Non-Dilution
8/20 kft Rapid Decompression
Absolute Pressure [mmHg]
600.0
500.0
400.0
300.0
200.0
100.0
0.0
-100.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

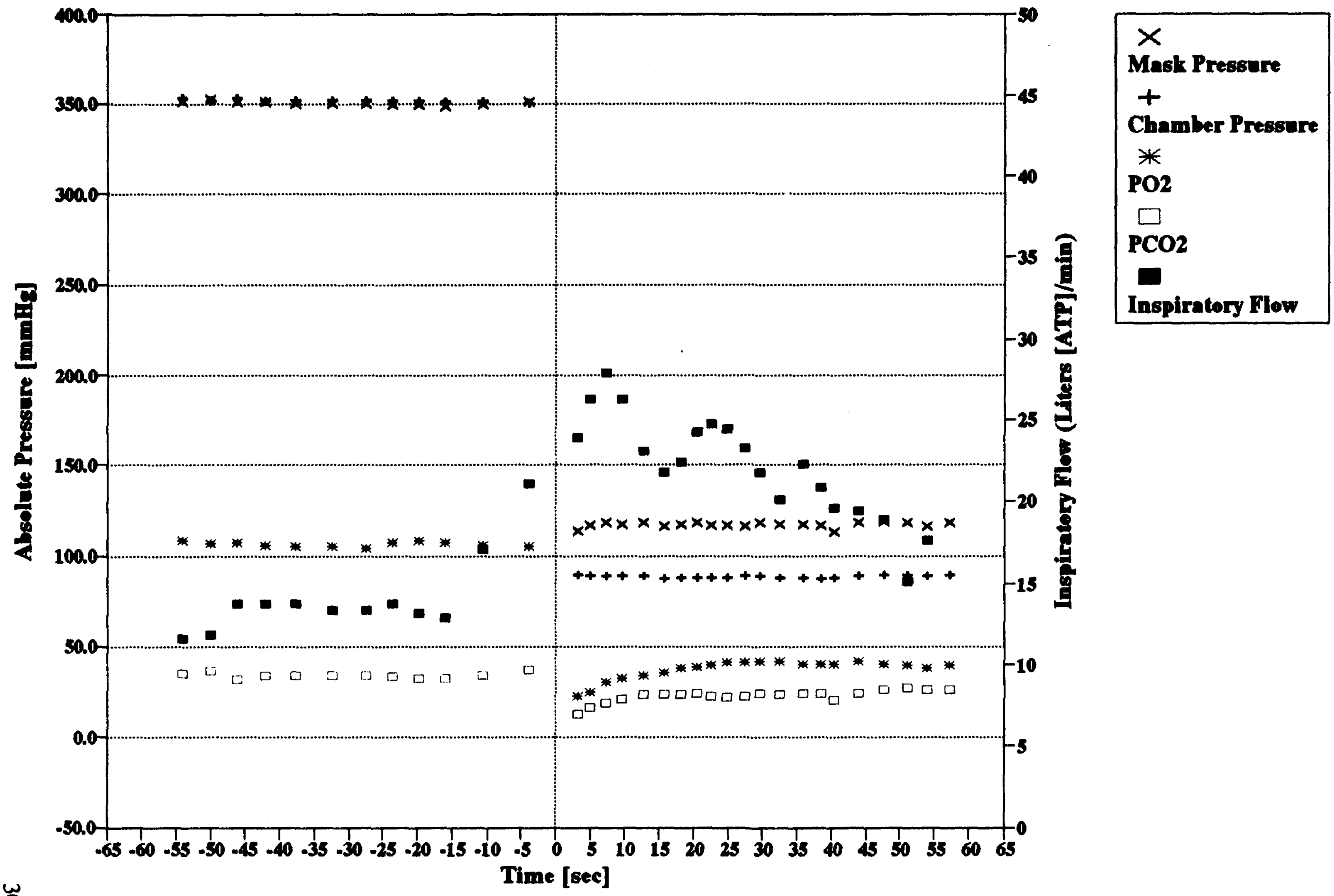


Subject: SCO / 100% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

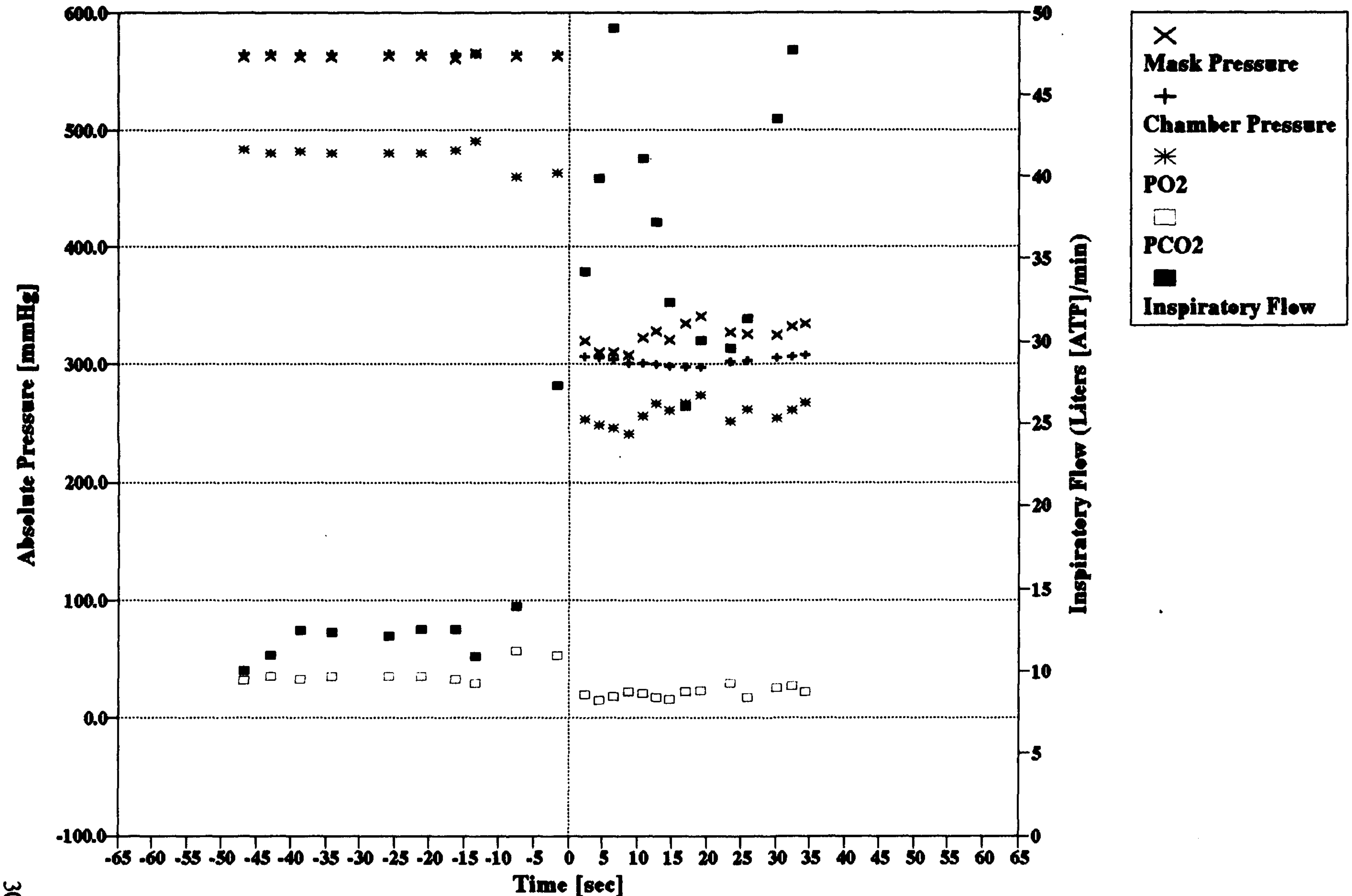

Subject: SCO / 100% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow


Subject: SCO / 93% O2 Non-Dilution
20/50 kft Rapid Decompression
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow
Inspiratory Flow (Liters [ATP]/min)
Absolute Pressure [mmHg]
Time [sec]

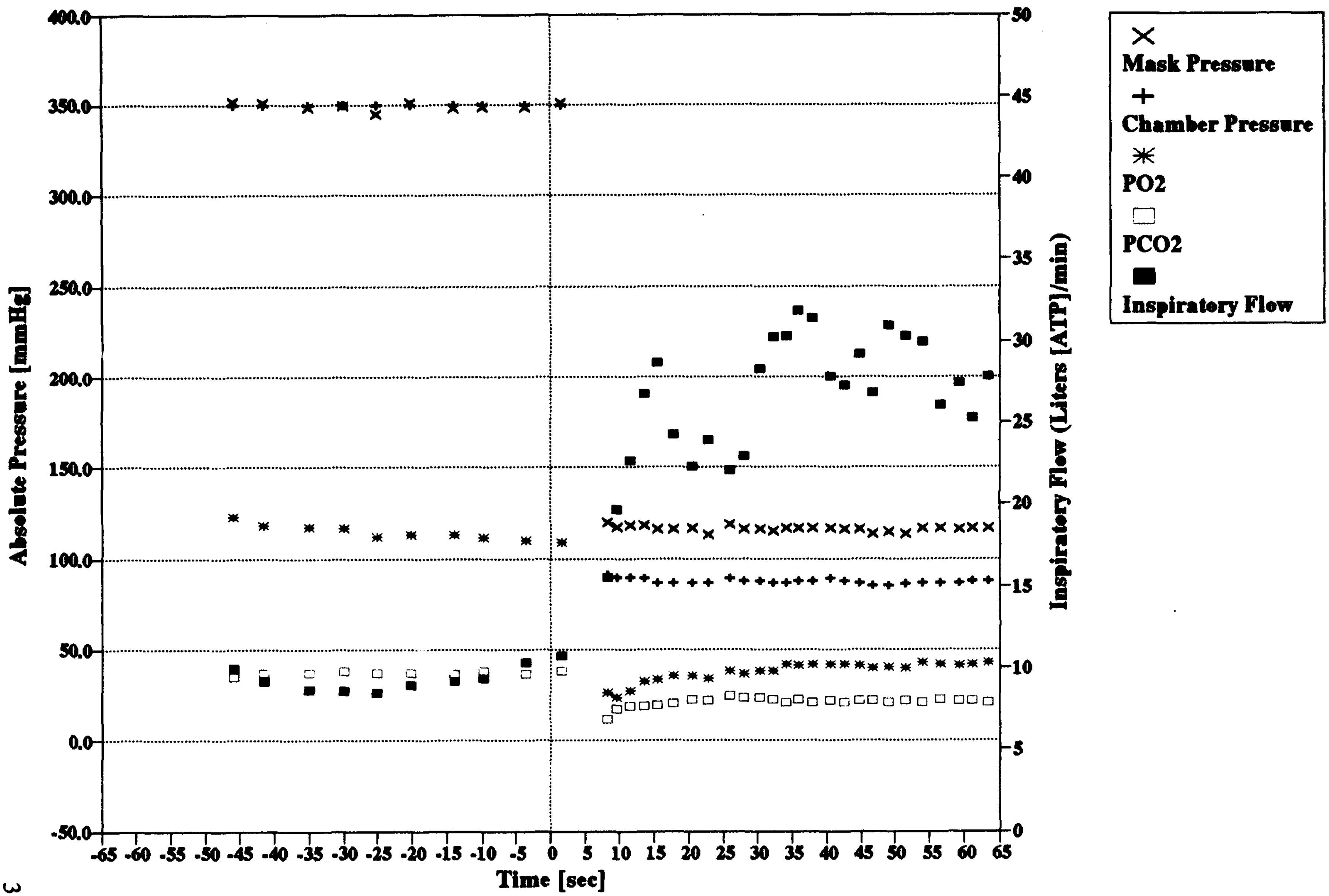
Subject: SCO / 93% O2 Dilution
20/50 kft Rapid Decompression

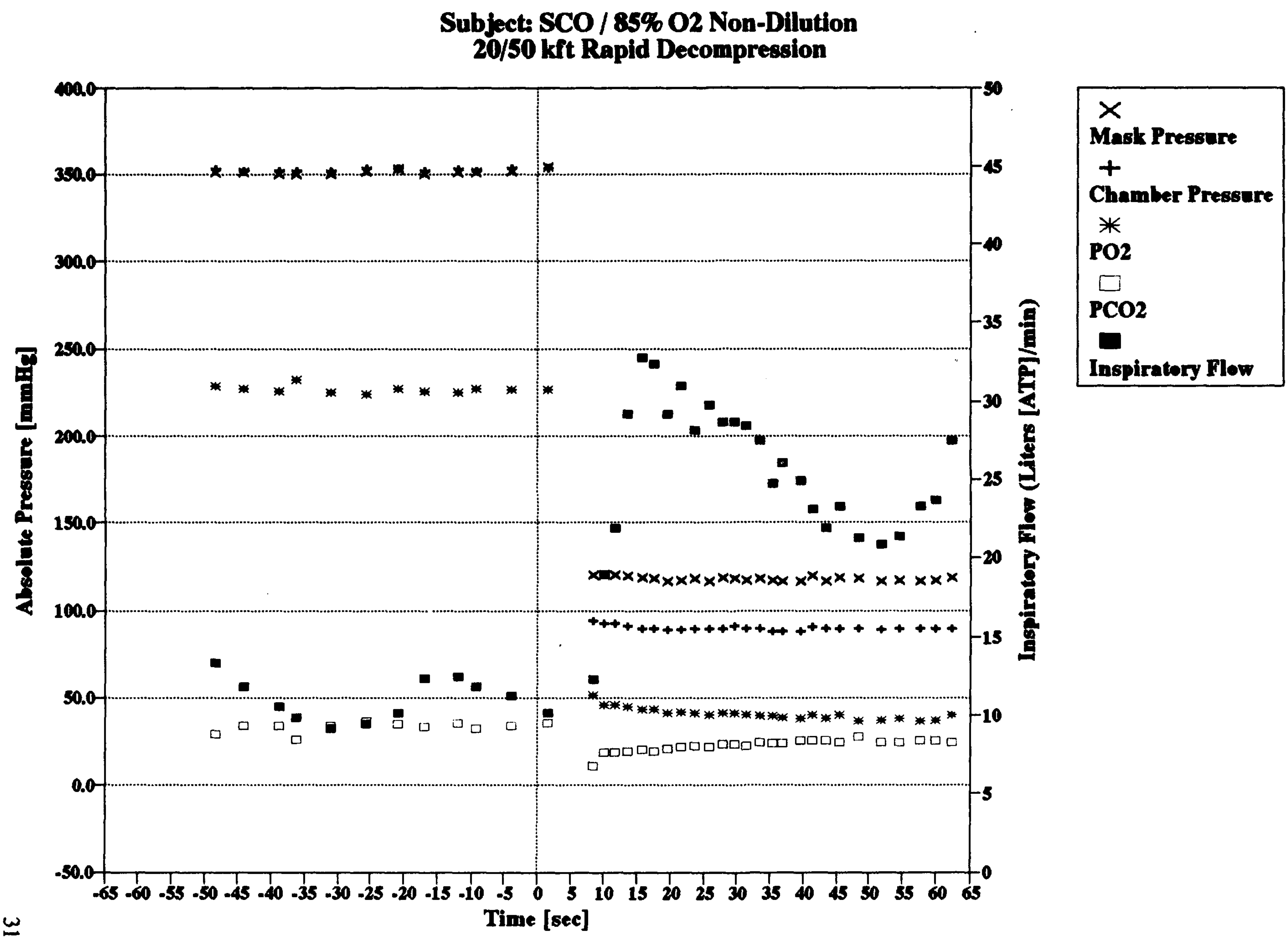


Subject: SCO / 90% O2 Non-Dilution
8/20 kft Rapid Decompression



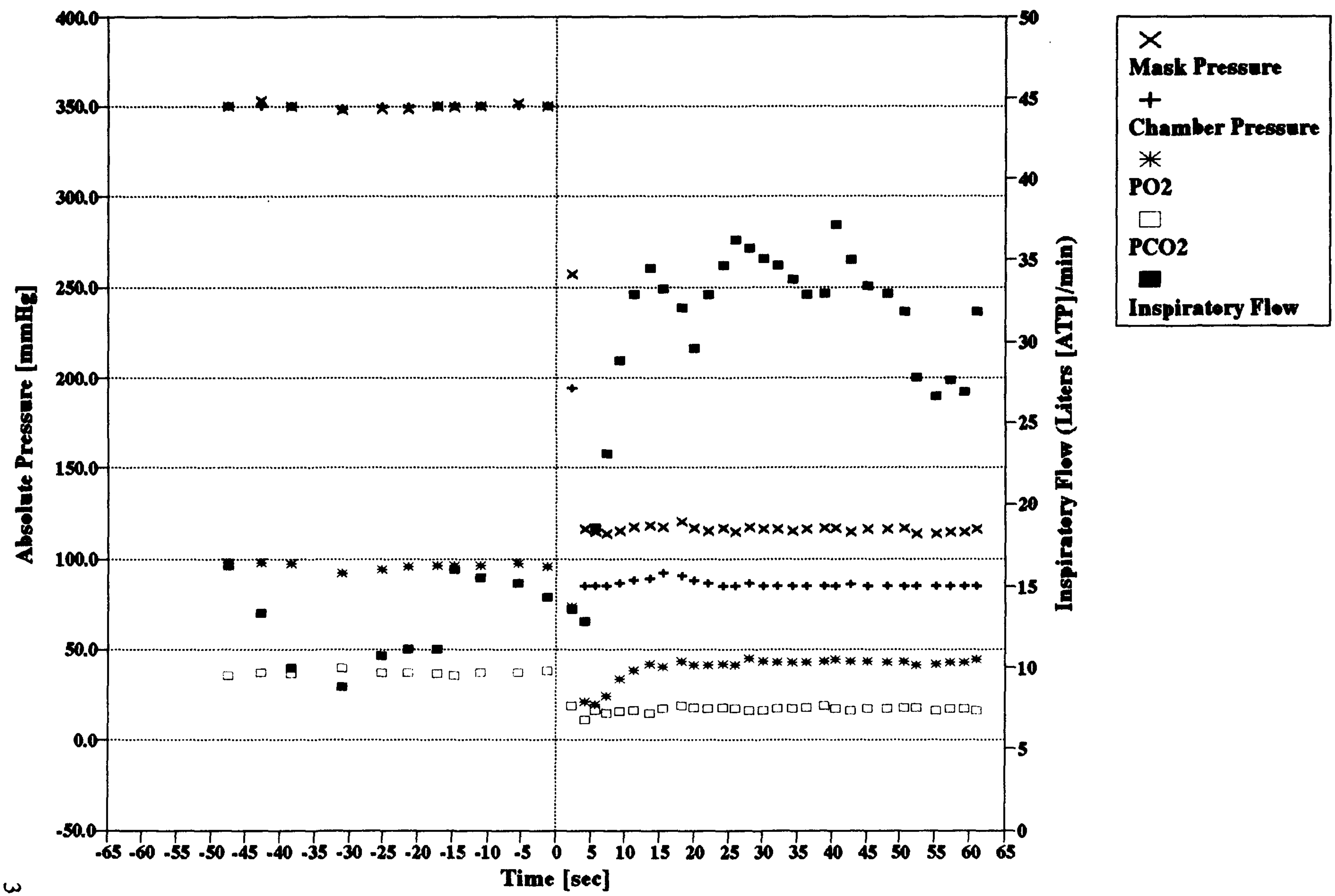
Subject: SCO / 90% O2 Dilution
20/50 kft Rapid Decompression



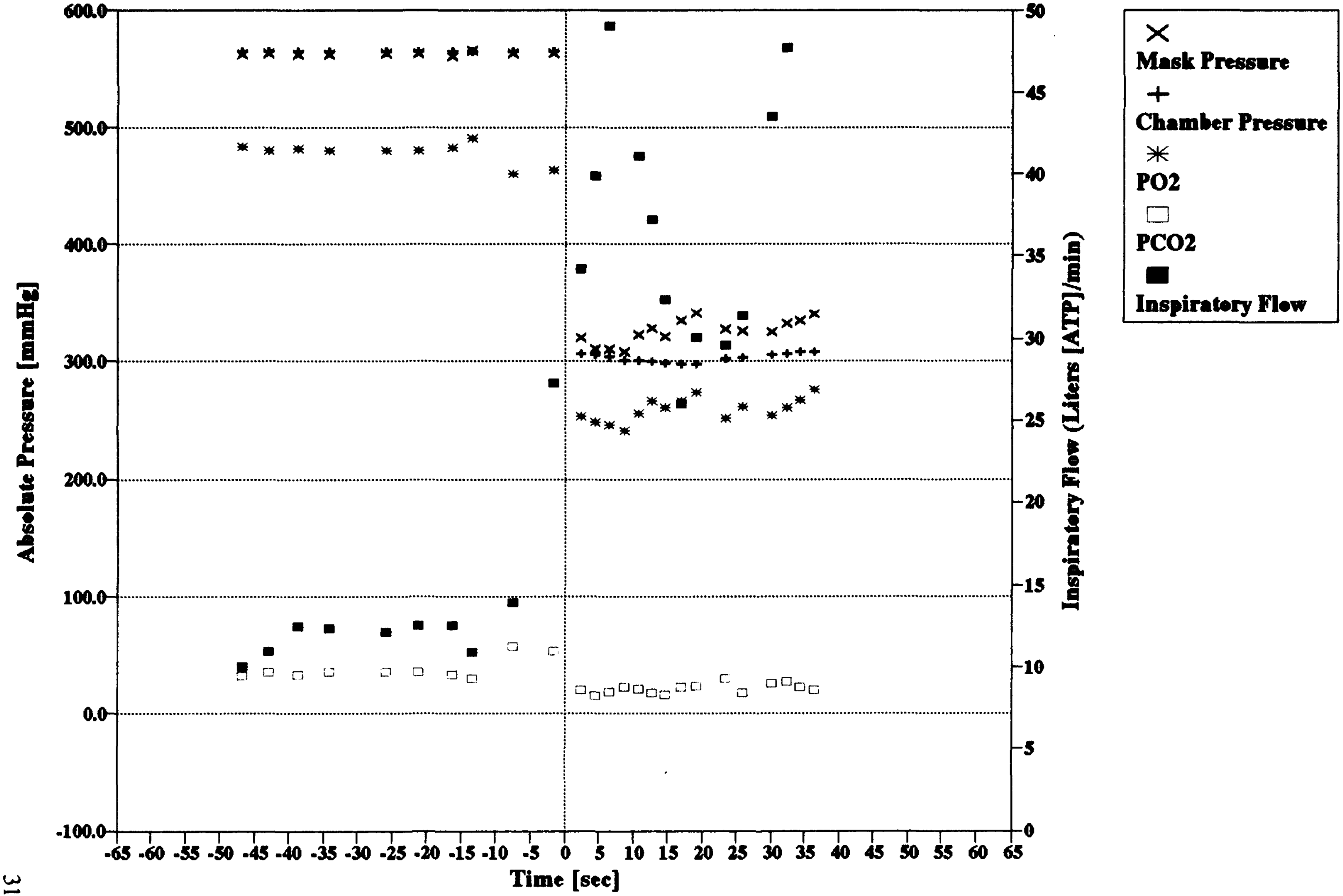


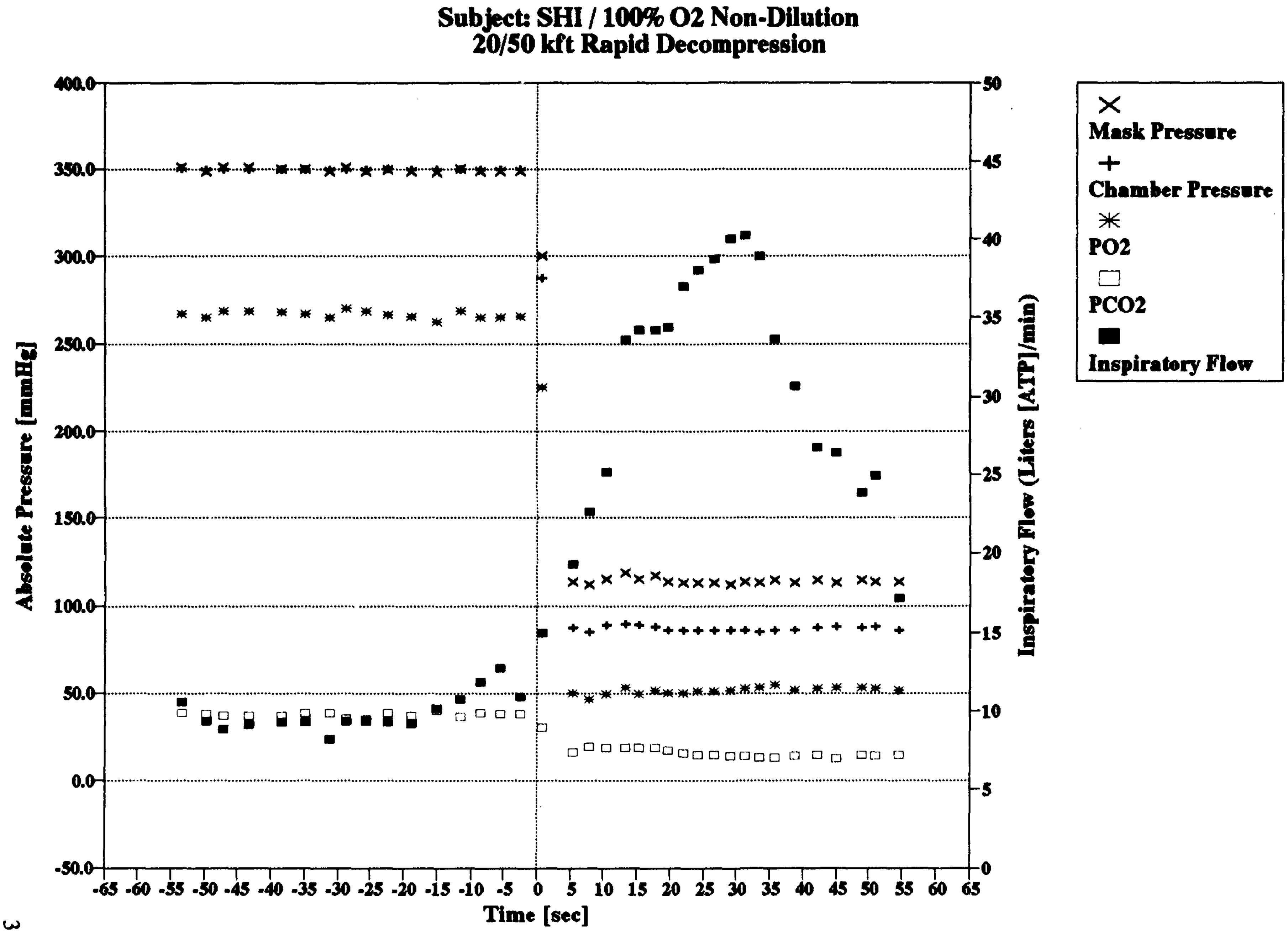
Subject: SCO / 85% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: SCO / 85% O2 Dilution
20/50 kft Rapid Decompression



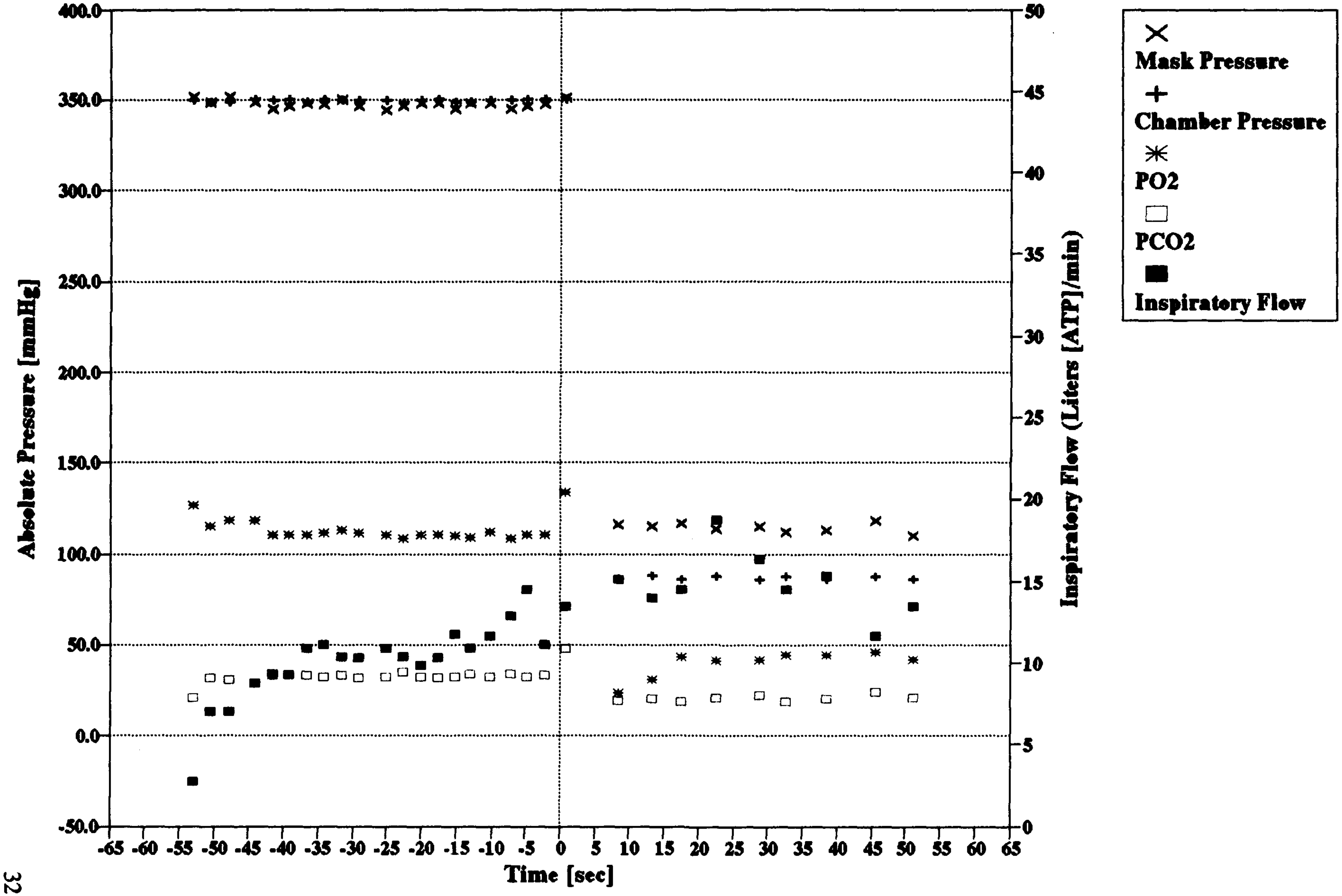
Subject: SCO / 100% O2 Non-Dilution
8/20 kft Rapid Decompression



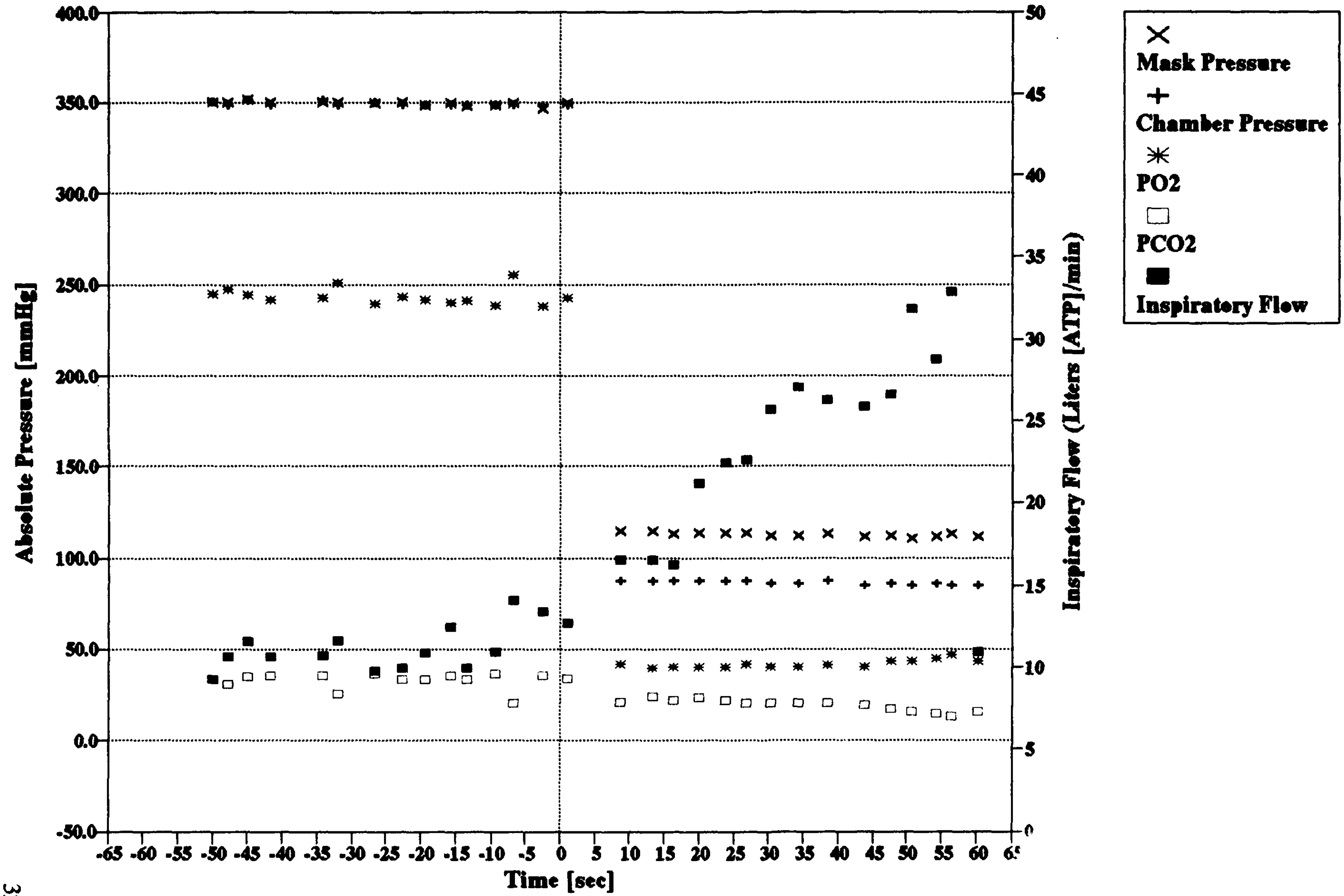


Subject: SHI / 100% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

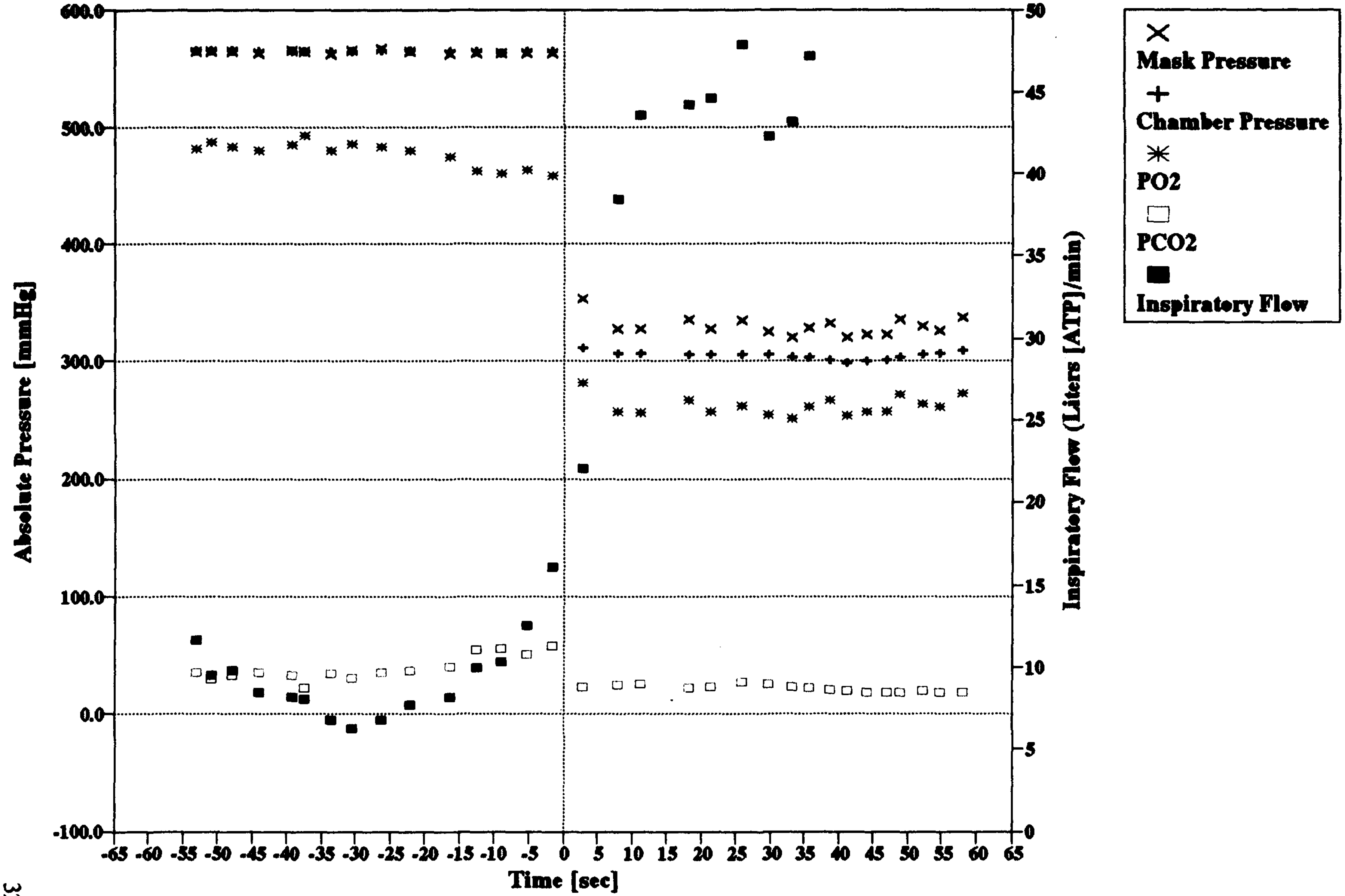
Subject: SHI / 100% O2 Dilution
20/50 kft Rapid Decompression



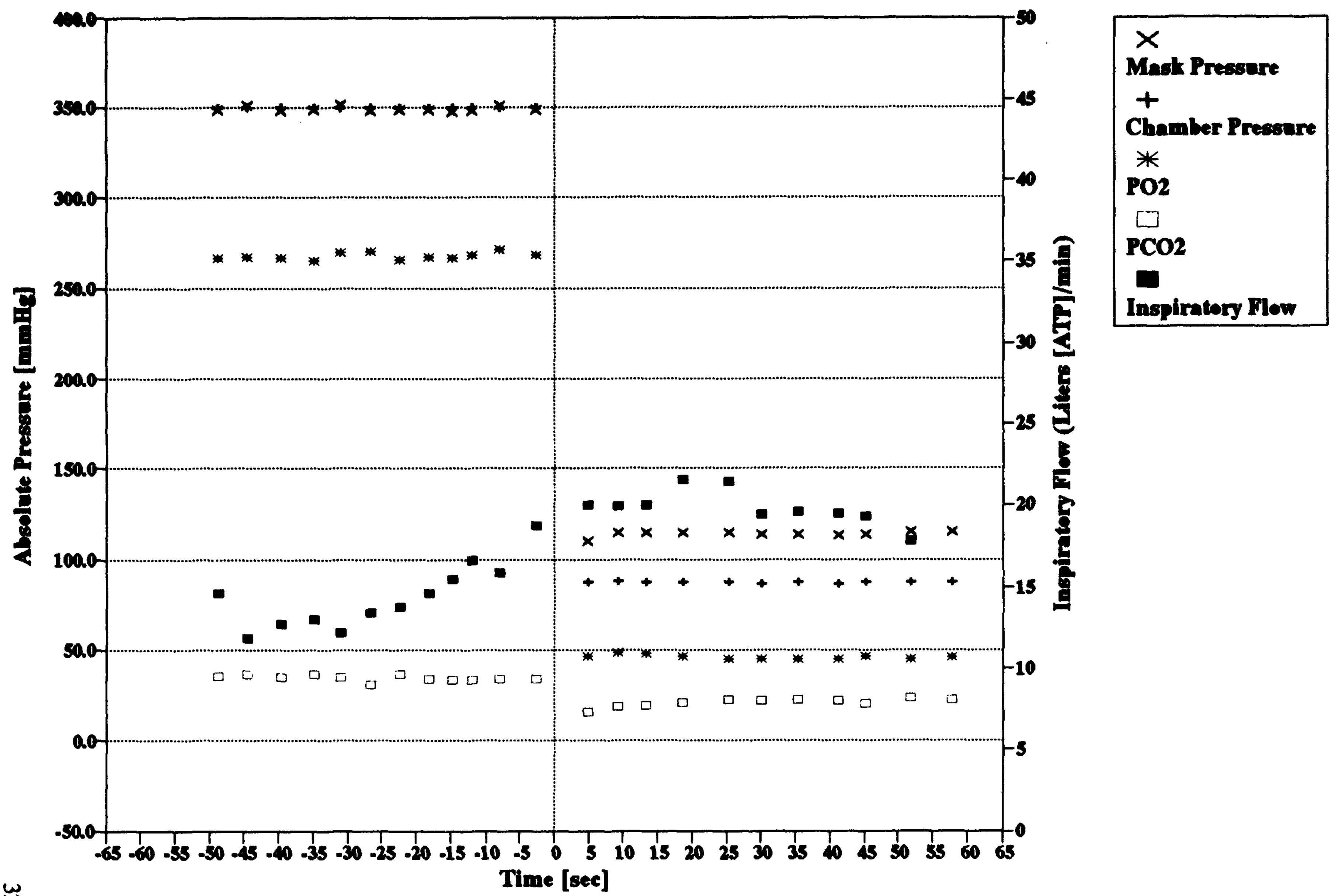
Subject: SHI / 90% O2 Non-Dilution
20/50 kft Rapid Decompression



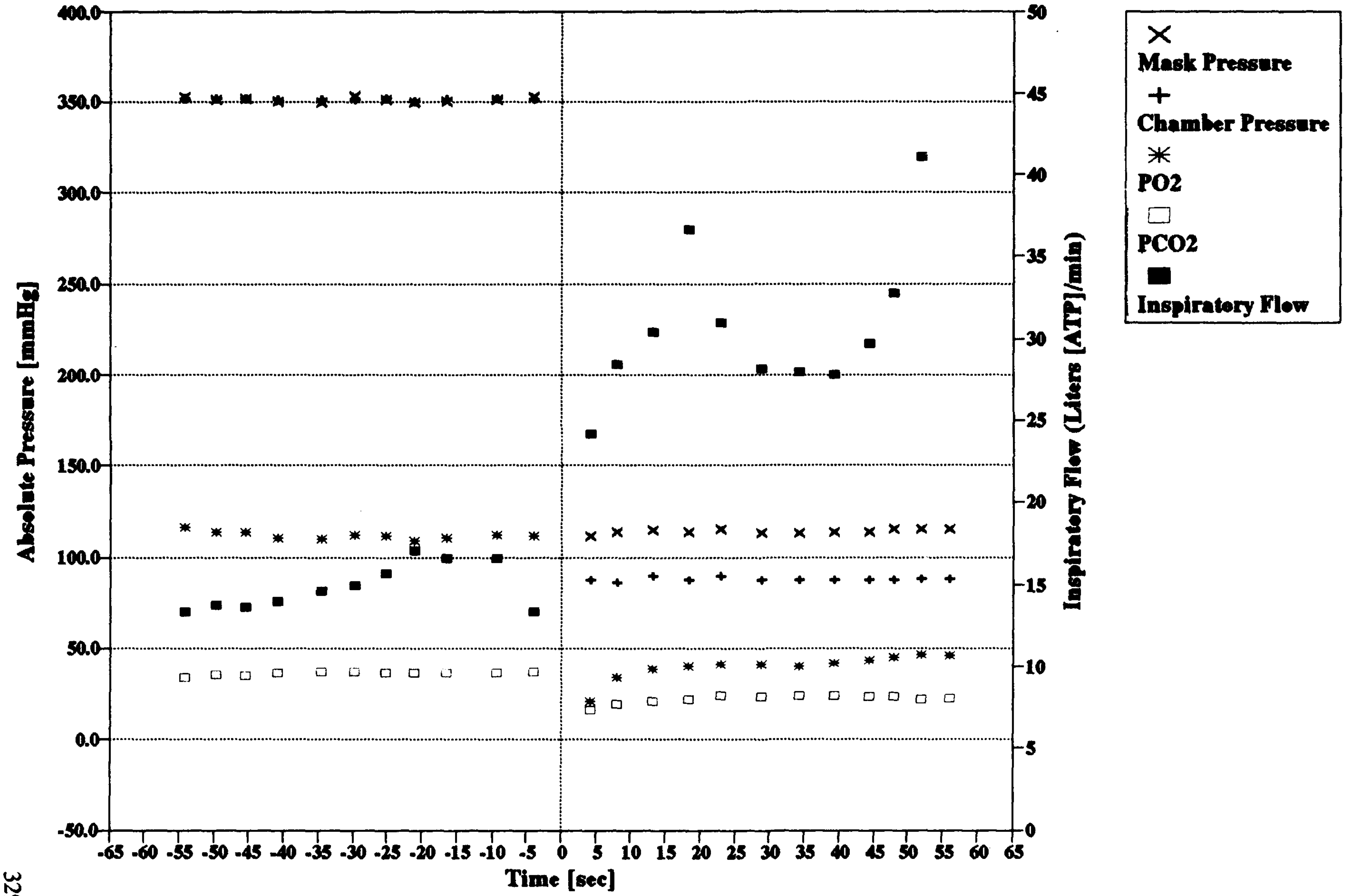
Subject: SHI / 100% O2 Non-Dilution
8/20 kft Rapid Decompression



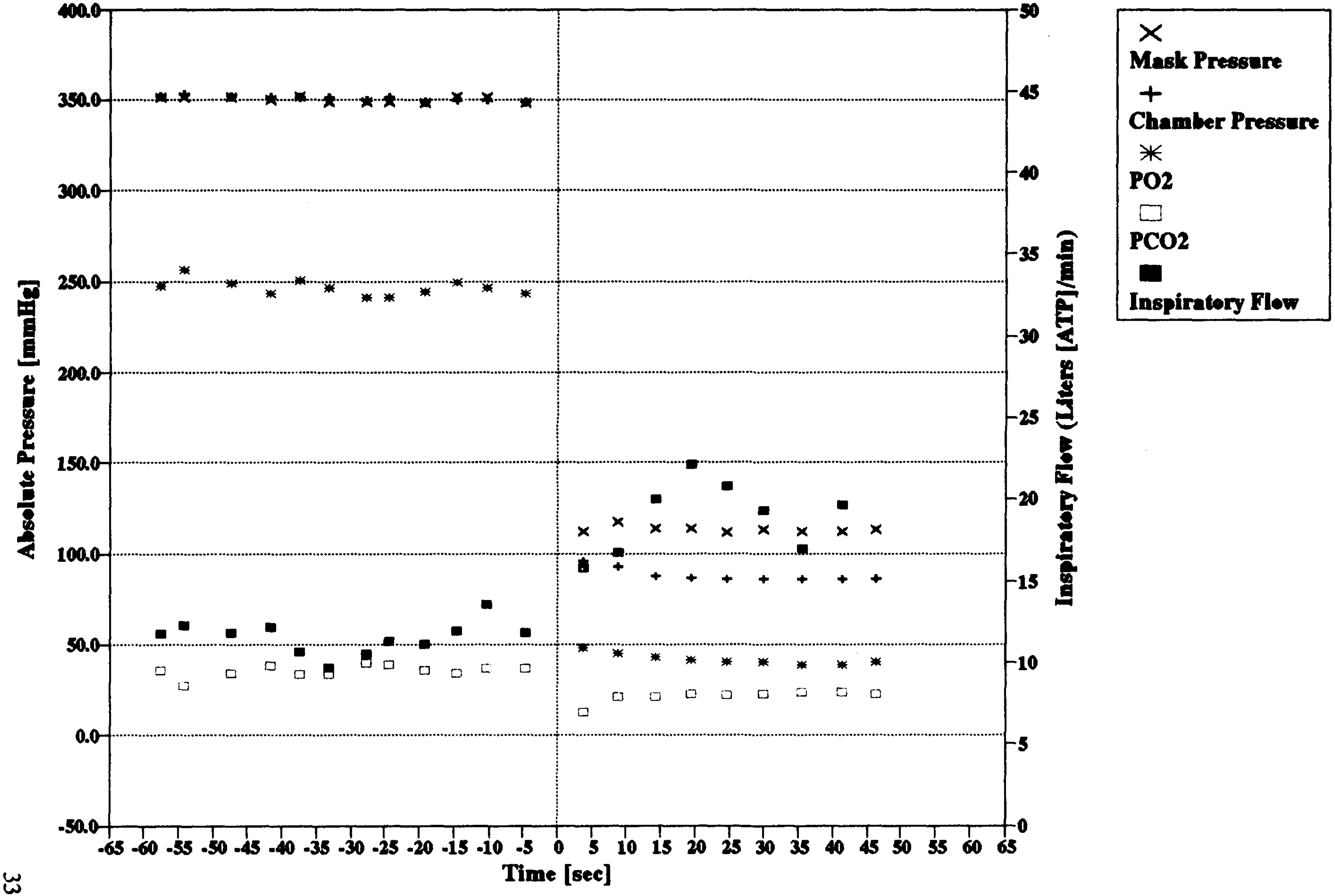
Subject: SHN / 100% O2 Non-Dilution
20/50 kft Rapid Decompression



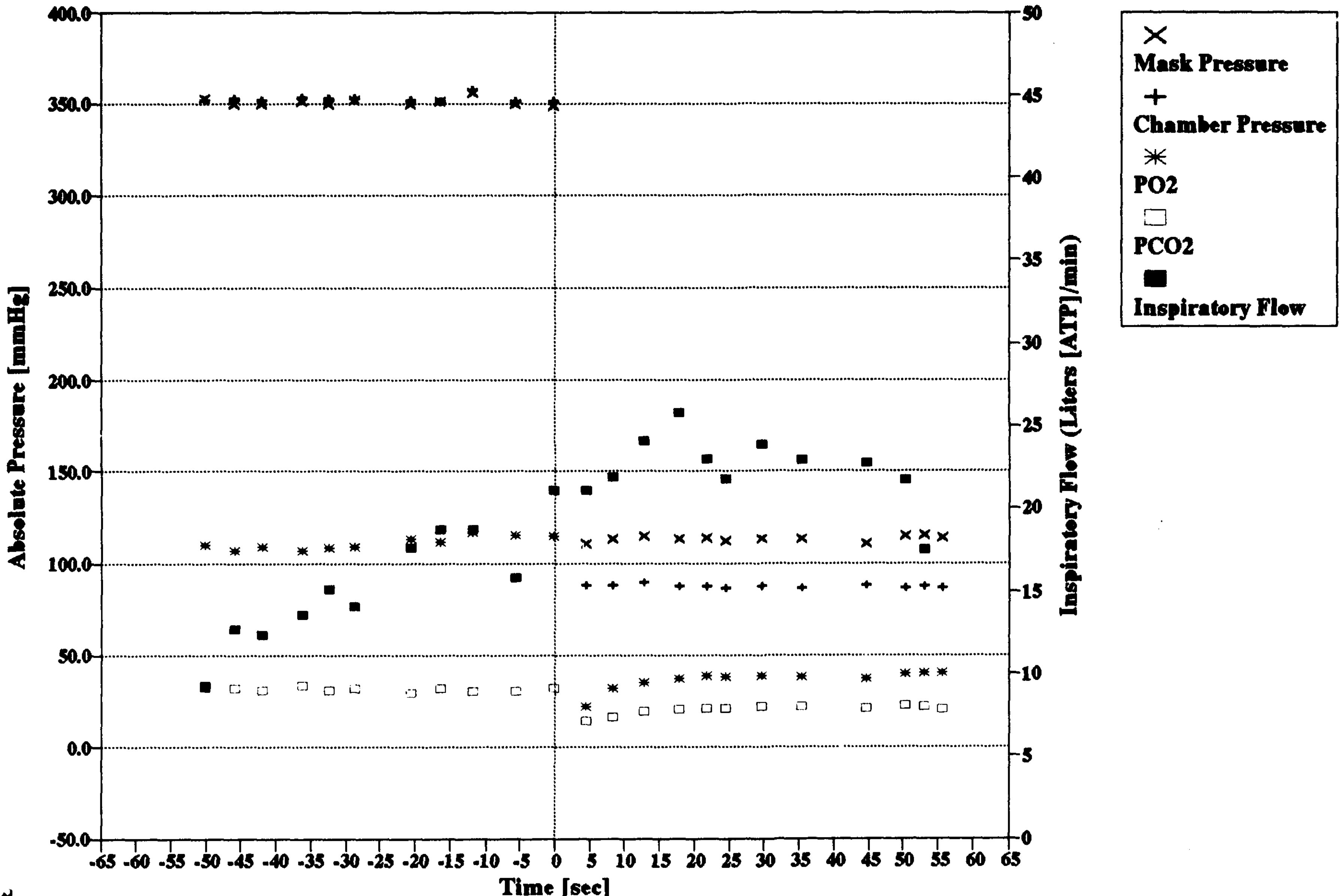
Subject: SHN / 100% O2 Dilution
20/50 kft Rapid Decompression



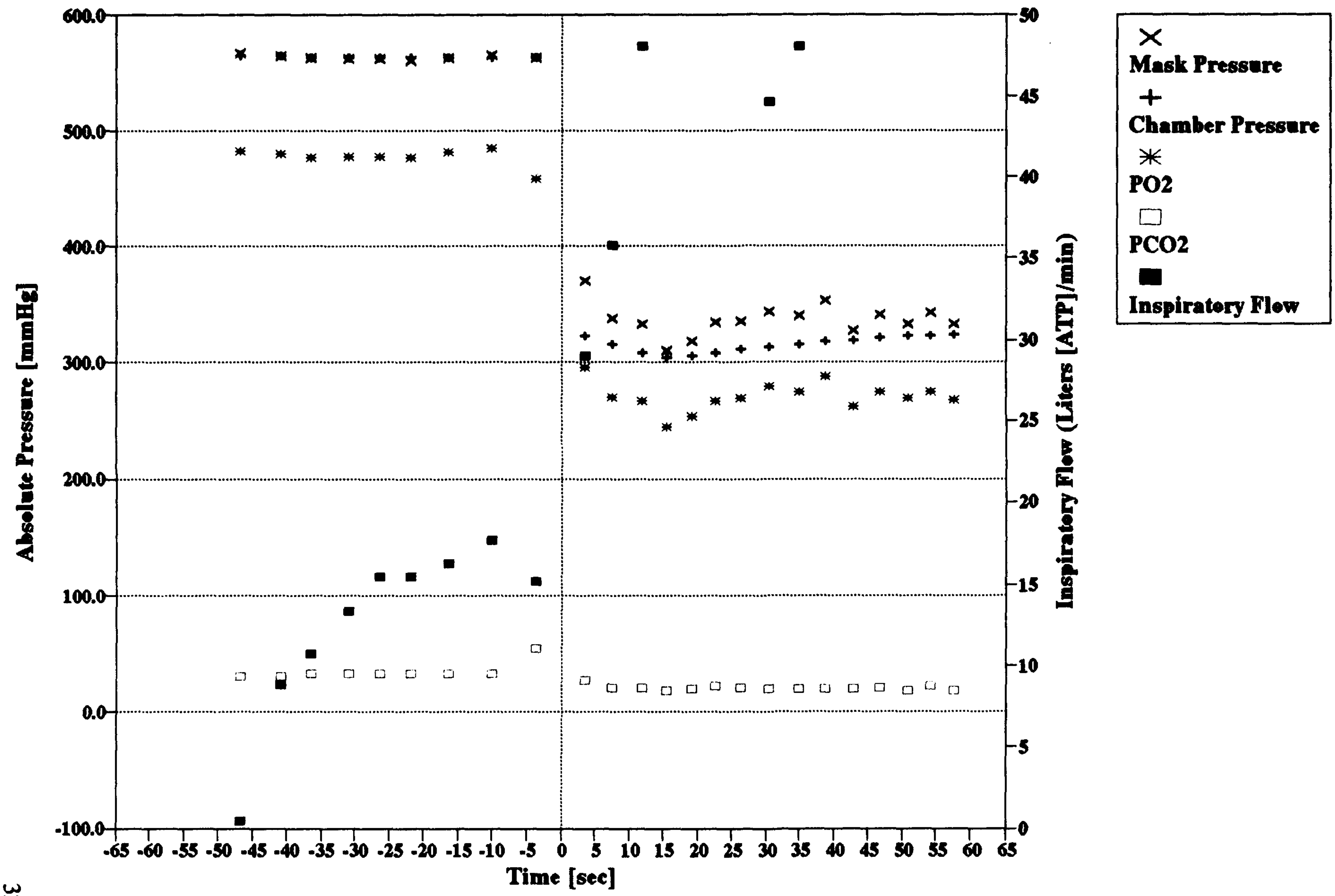
Subject: SHN / 93% O2 Non-Dilution
20/50 kft Rapid Decompression

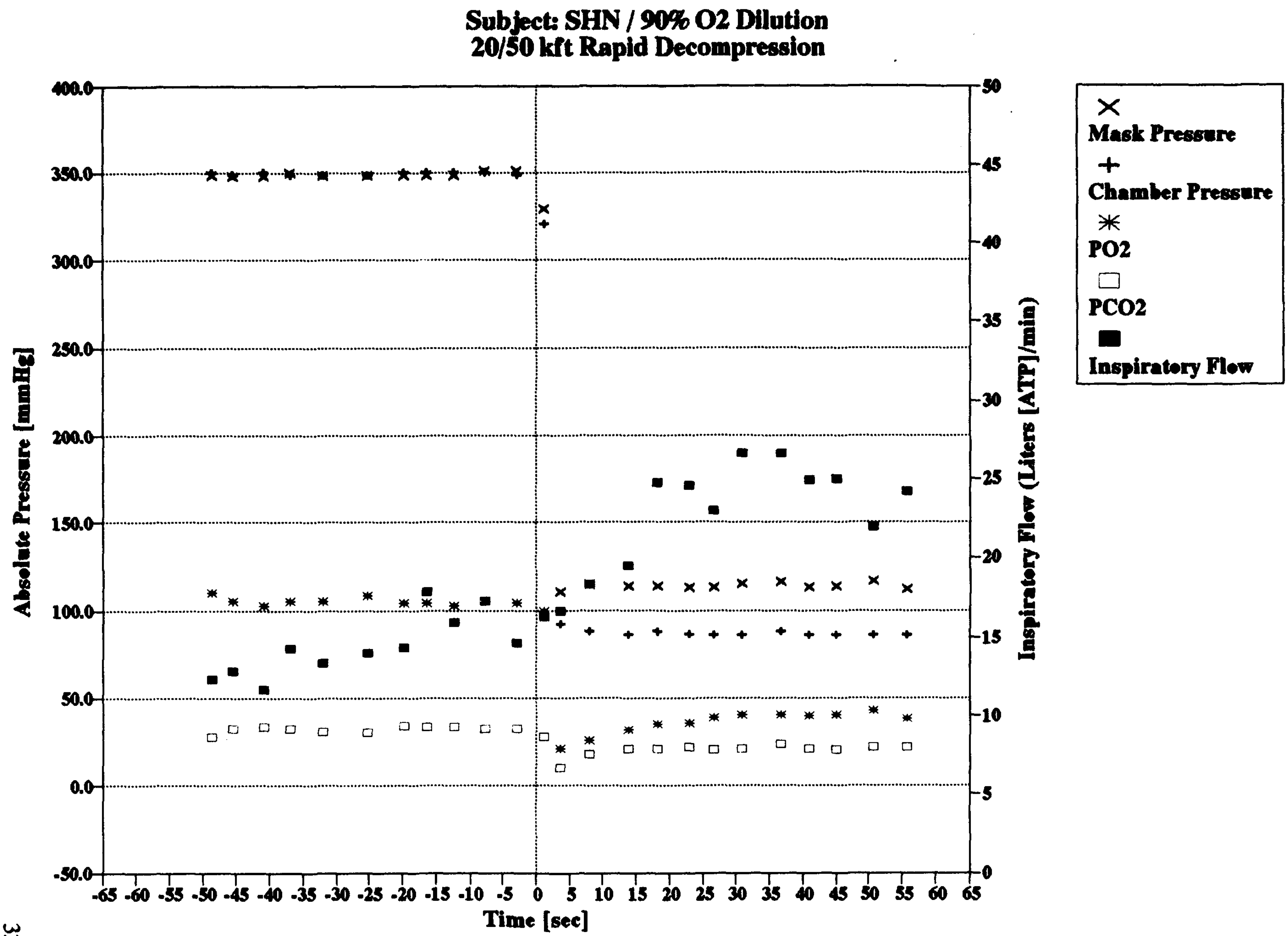


Subject: SHN / 93% O2 Dilution
20/50 kft Rapid Decompression



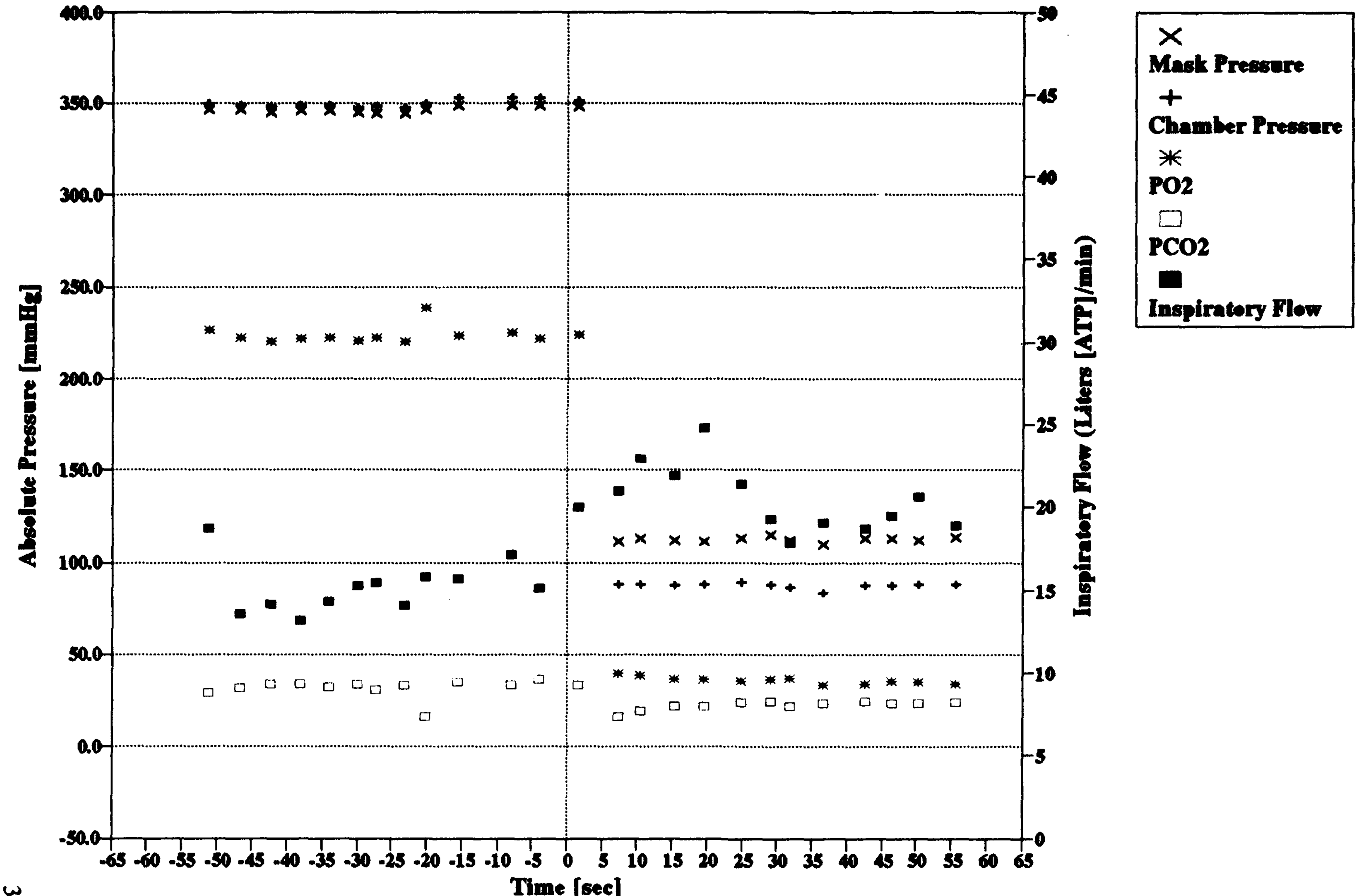
Subject: SHN / 90% O2 Non-Dilution
8/20 kft Rapid Decompression



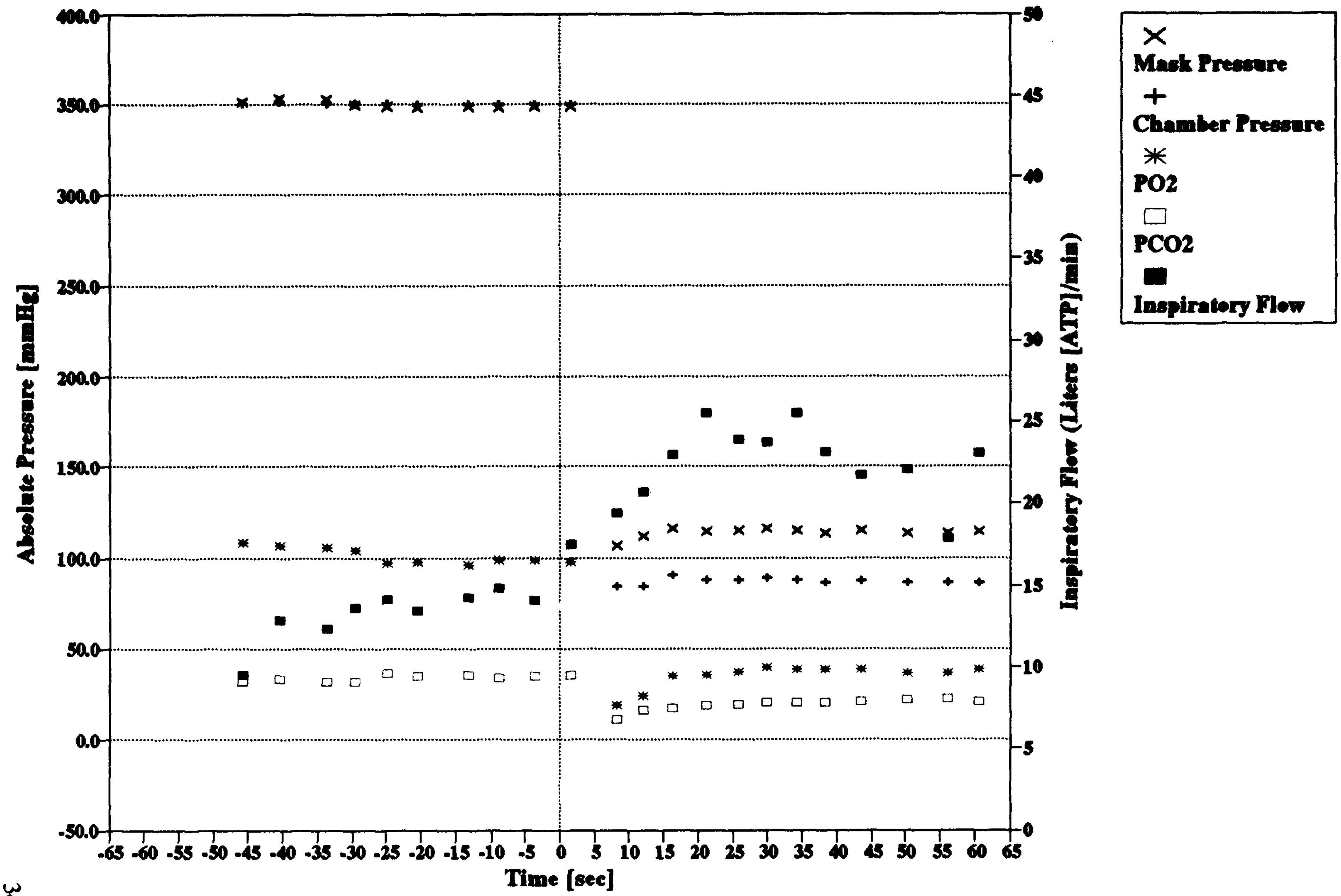


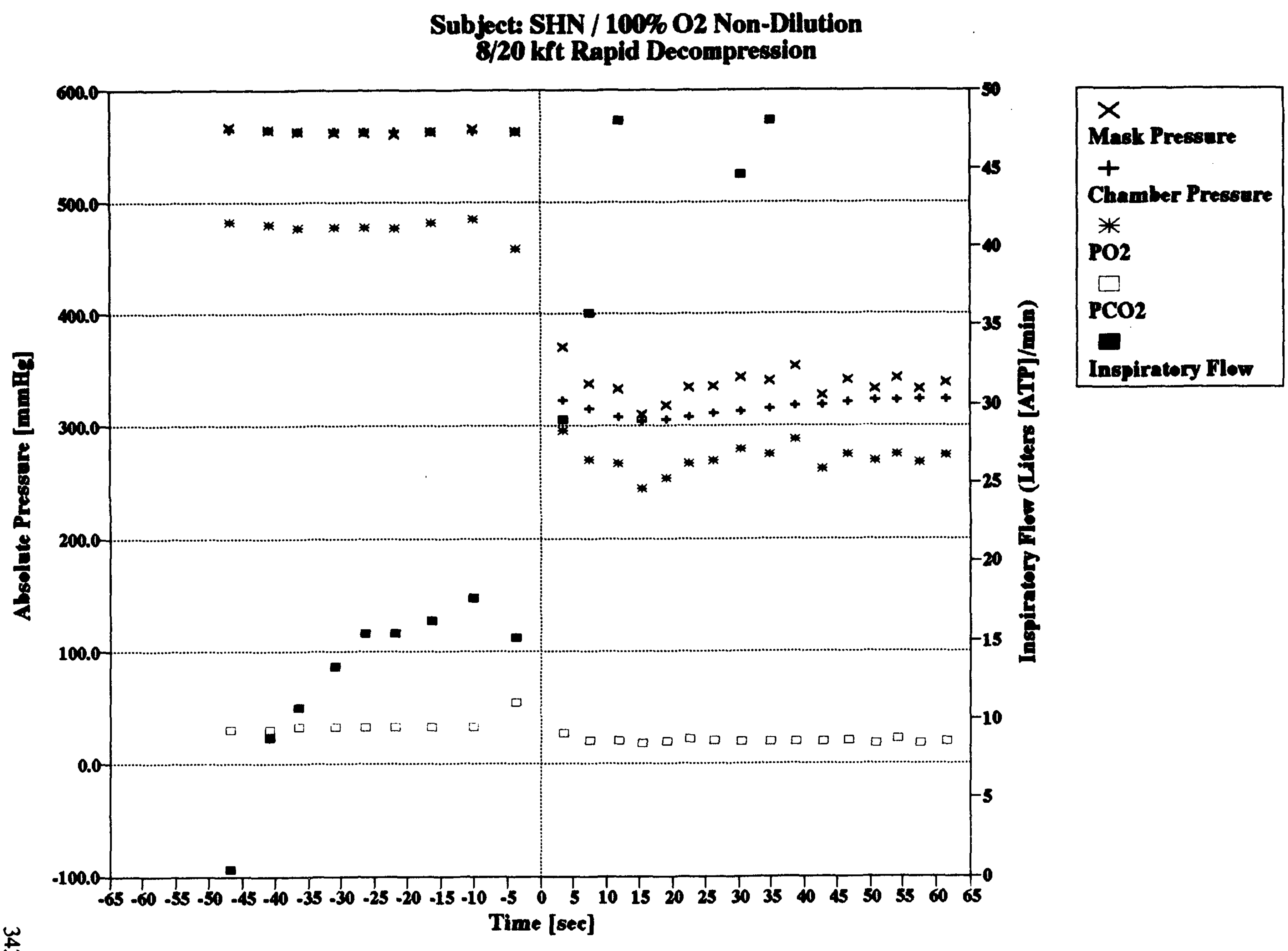
Subject: SHN / 90% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: SHN / 85% O2 Non-Dilution
20/50 kft Rapid Decompression

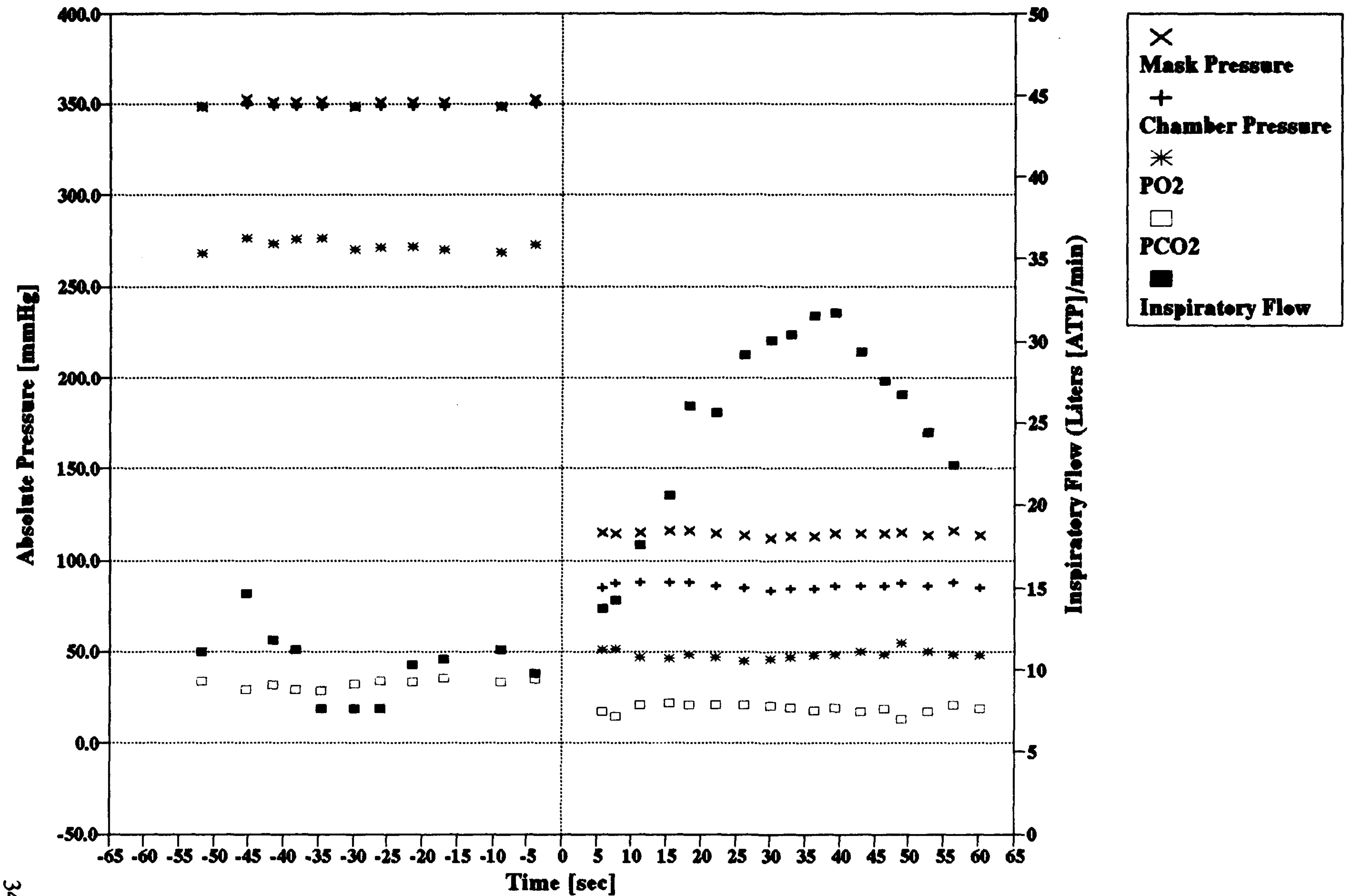


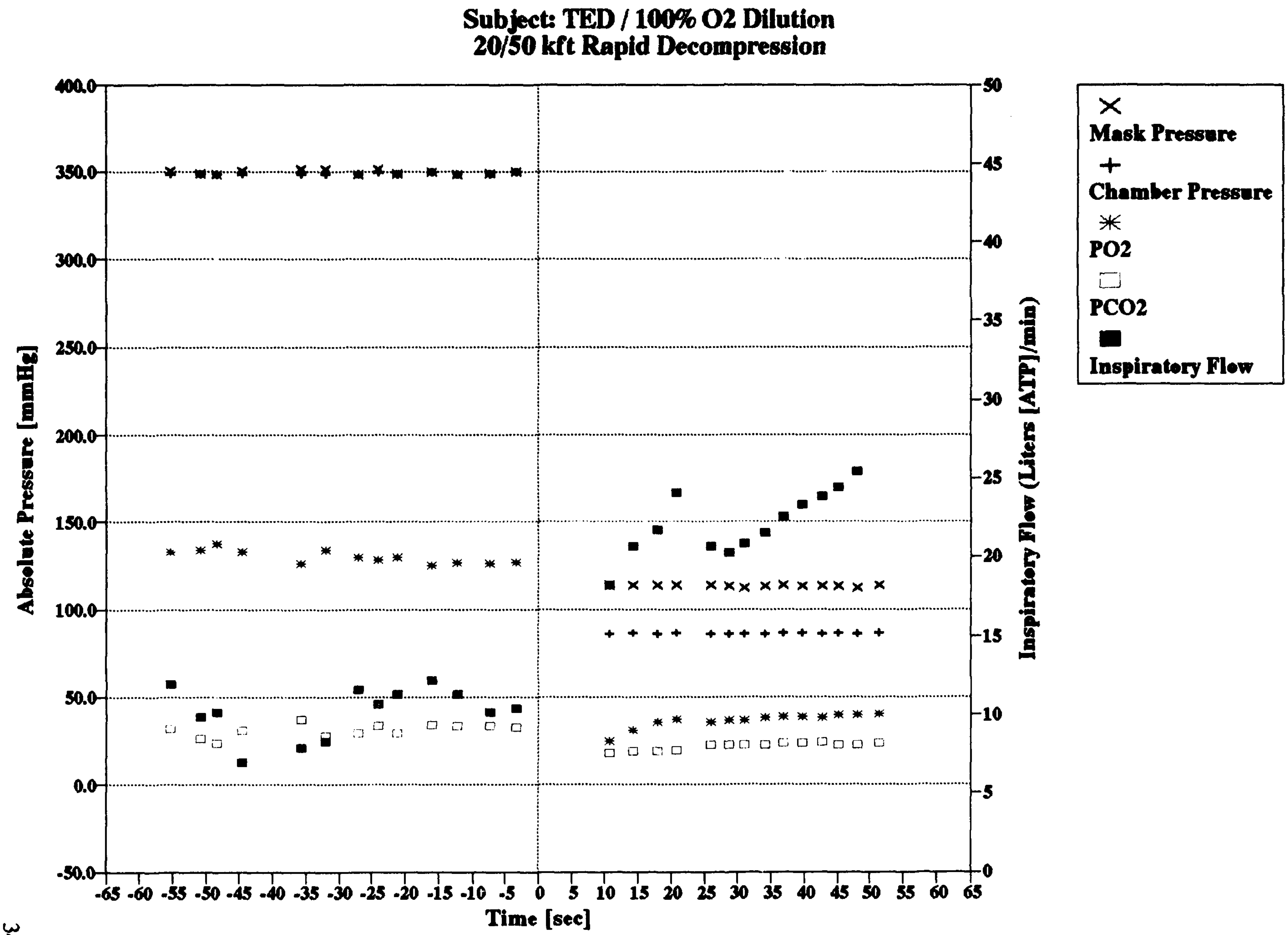
Subject: SHN / 85% O2 Dilution
20/50 kft Rapid Decompression



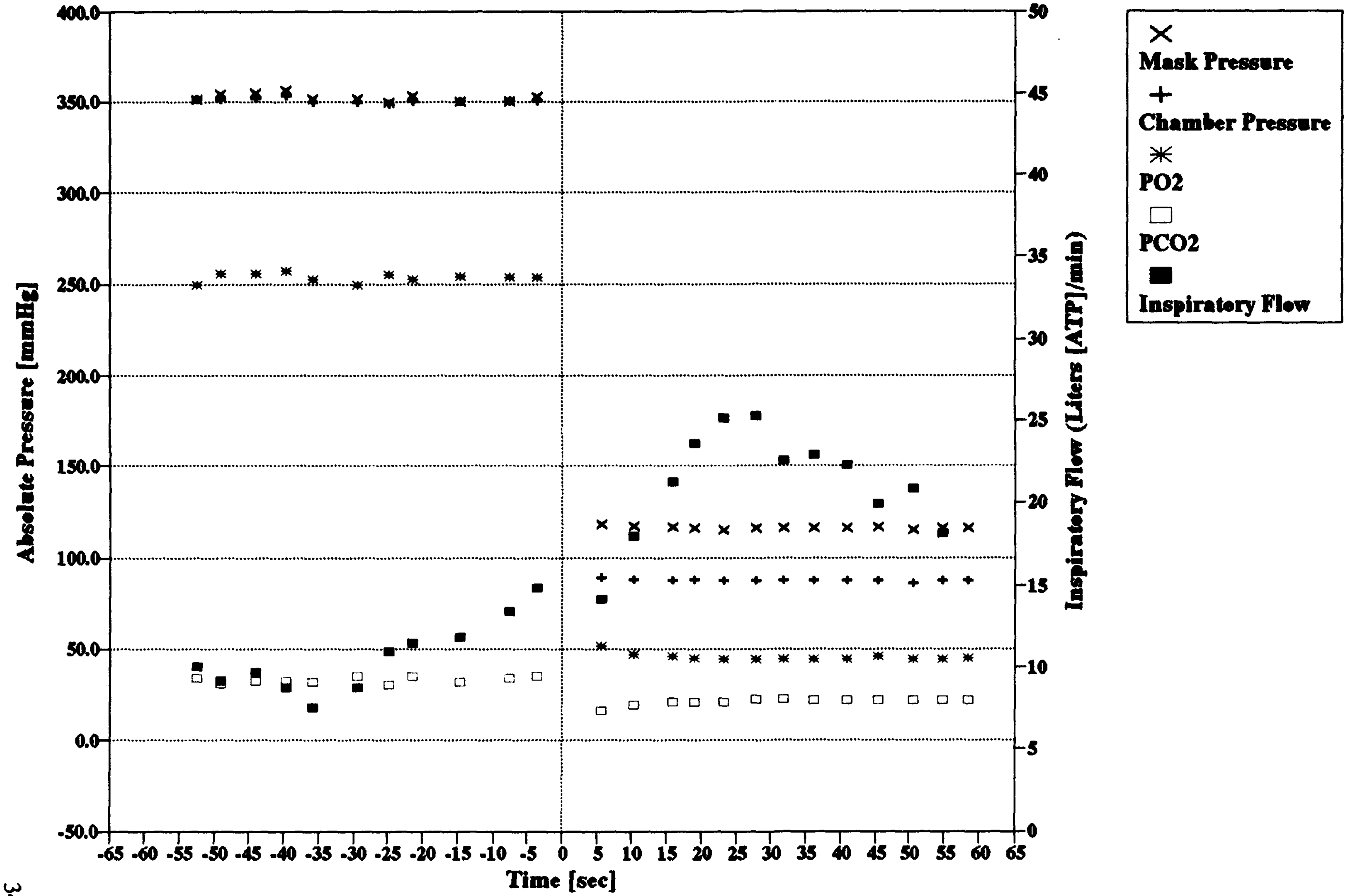

Subject: SHN / 100% O2 Non-Dilution
8/20 kft Rapid Decompression
Absolute Pressure [mmHg]
600.0
500.0
400.0
300.0
200.0
100.0
0.0
-100.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Subject: TED / 100% O2 Non-Dilution
20/50 kft Rapid Decompression

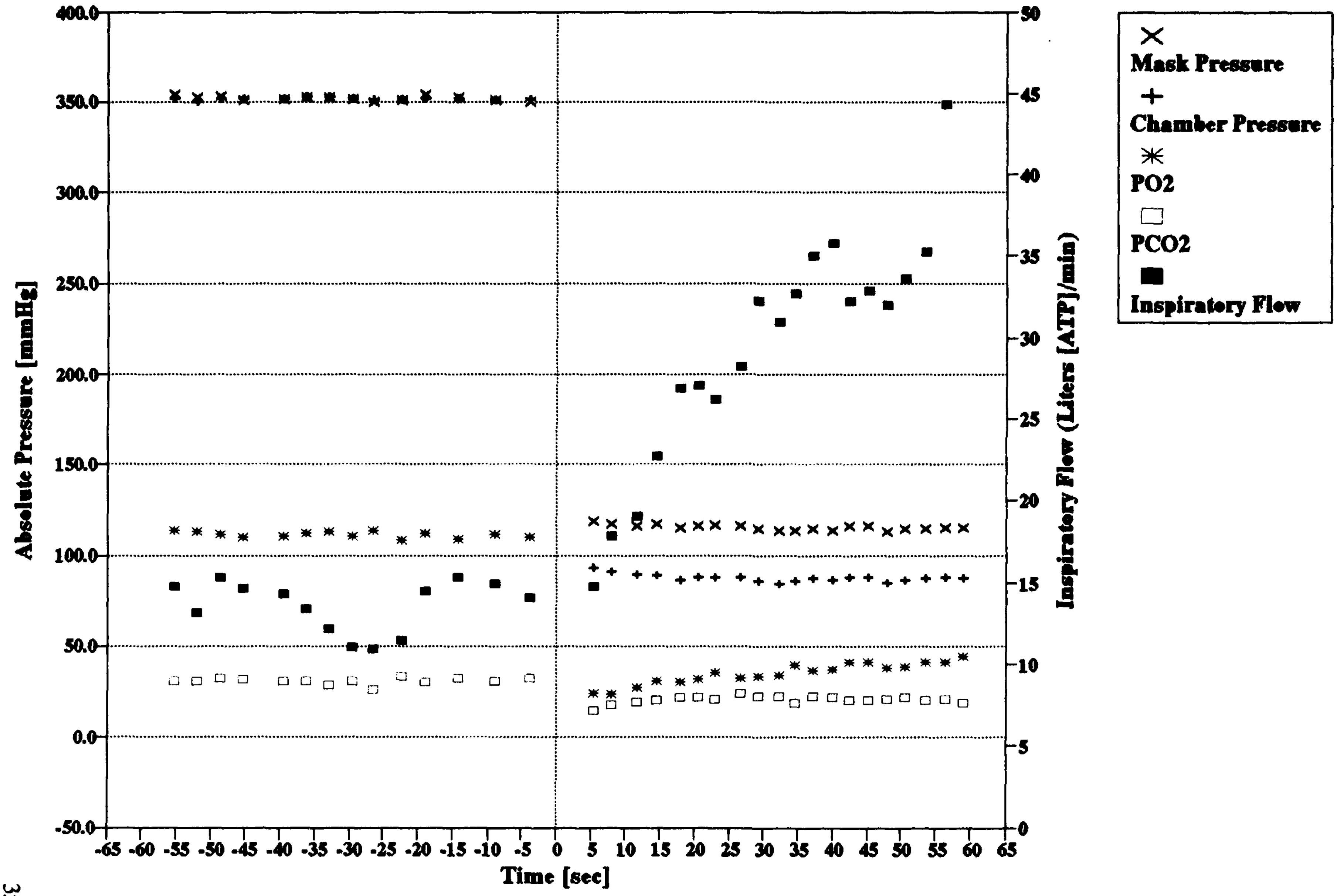



Subject: TED / 100% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

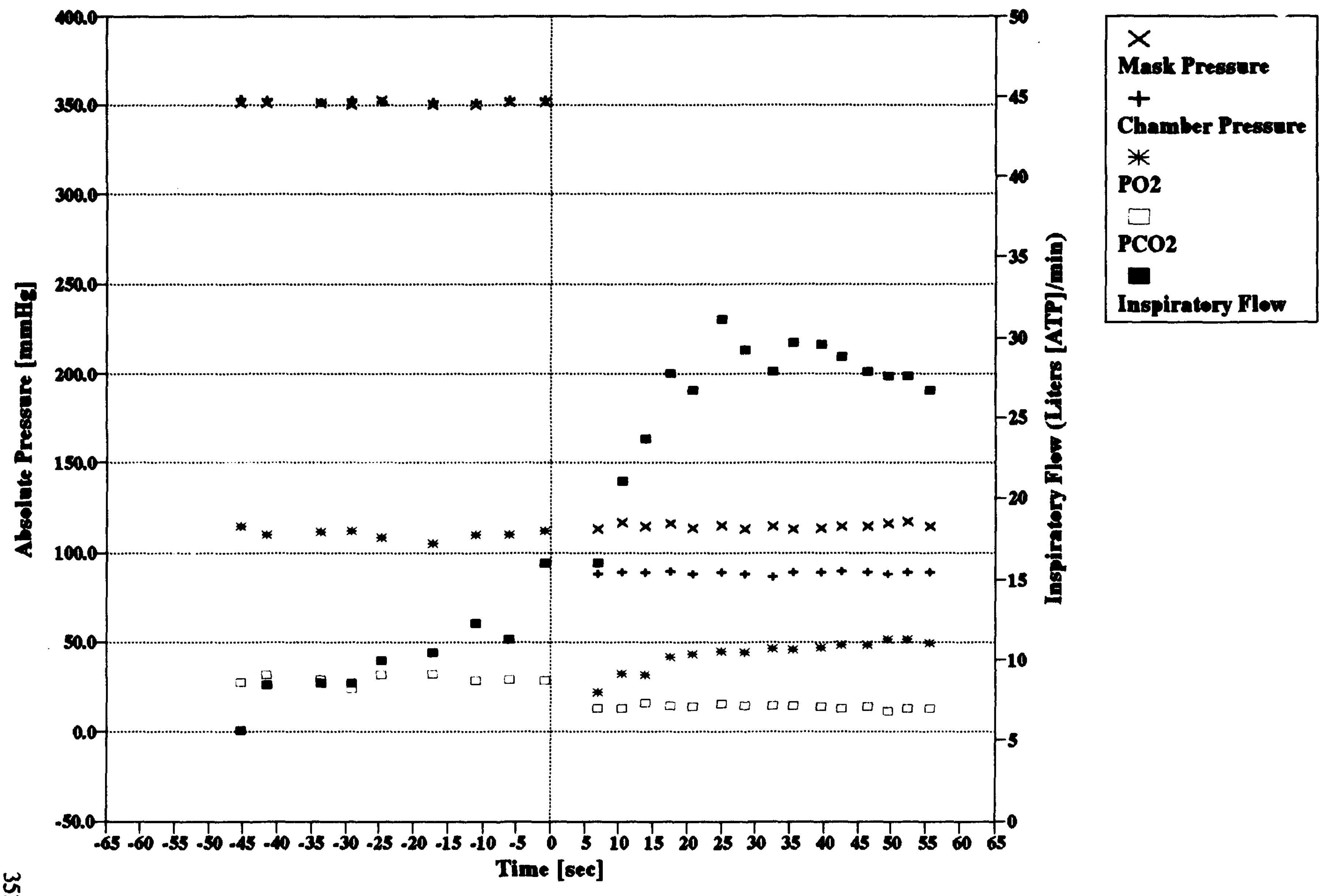
Subject: TED / 93% O2 Non-Dilution
20/50 kft Rapid Decompression



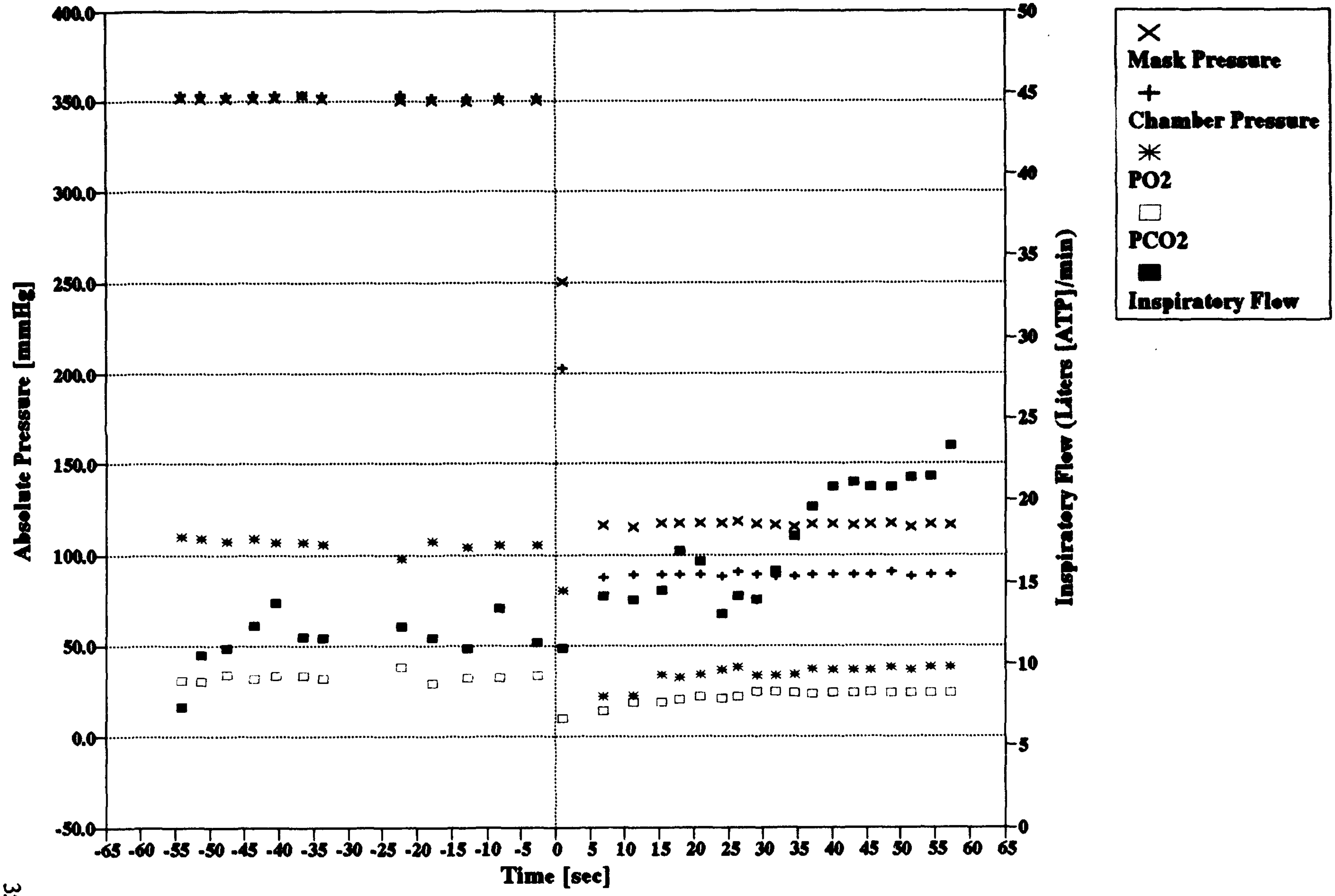
Subject: TED / 93% O2 Dilution
20/50 kft Rapid Decompression



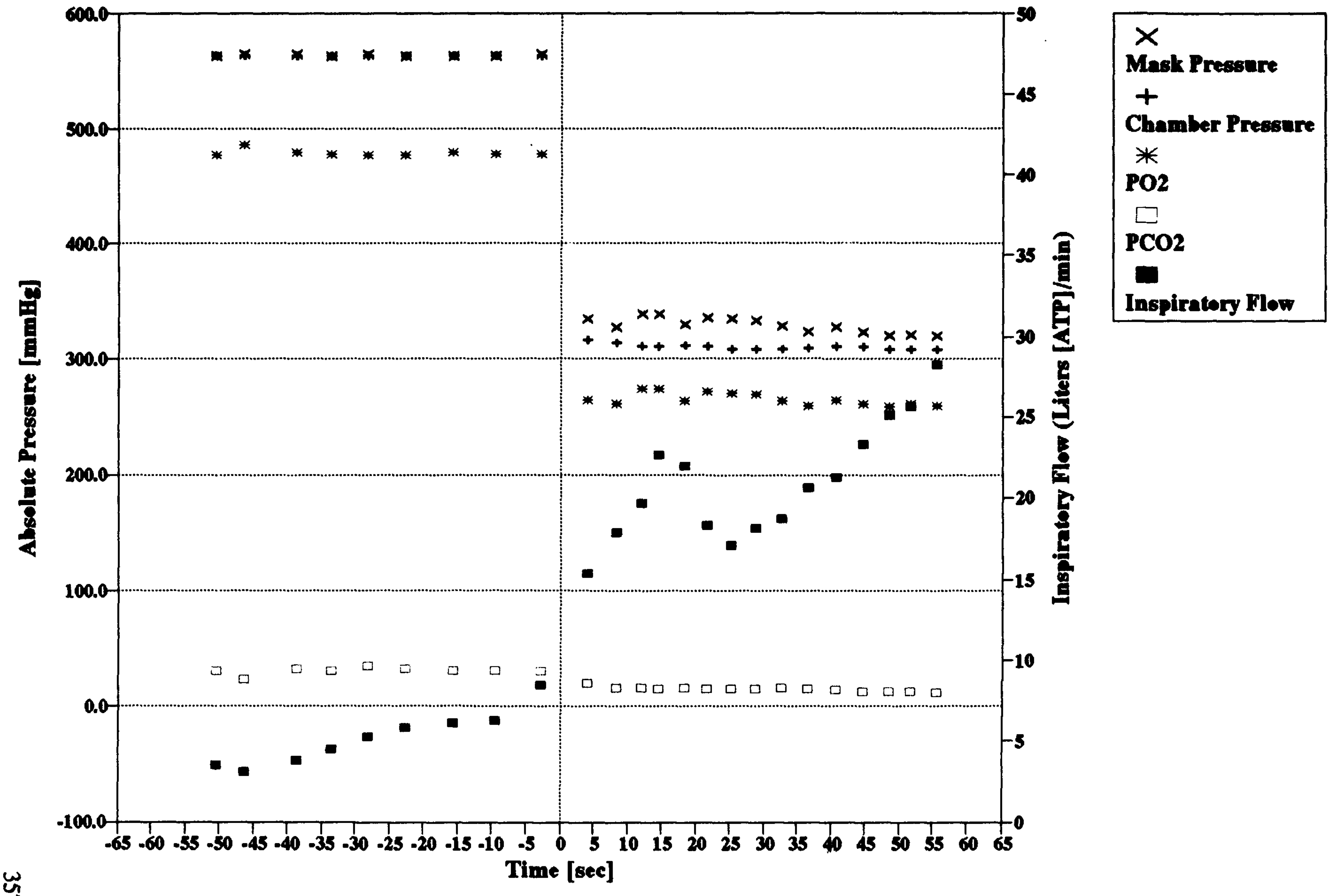
Subject: TED / 90% O2 Non-Dilution
20/50 kft Rapid Decompression



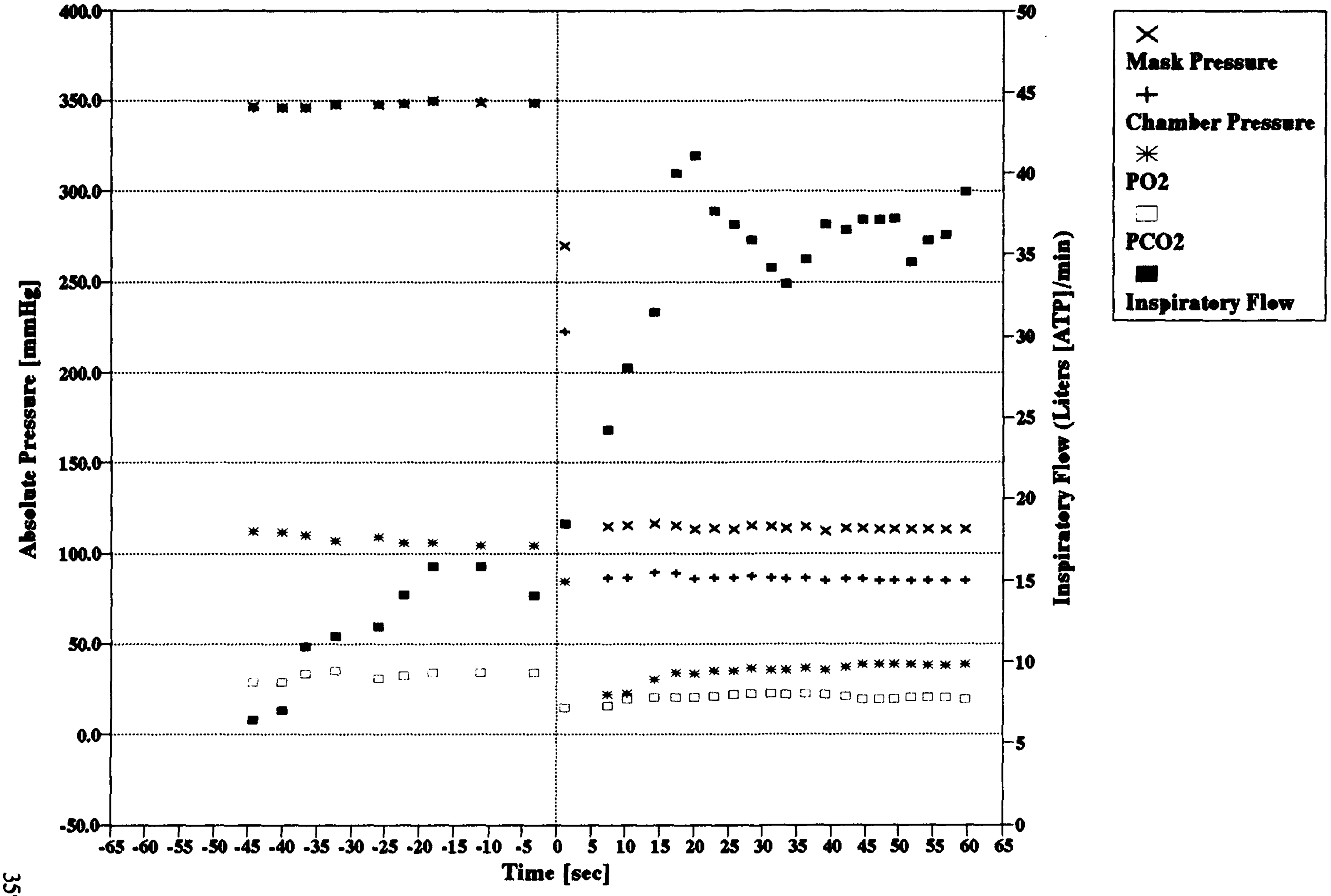
Subject: TED / 90% O2 Dilution
20/50 kft Rapid Decompression



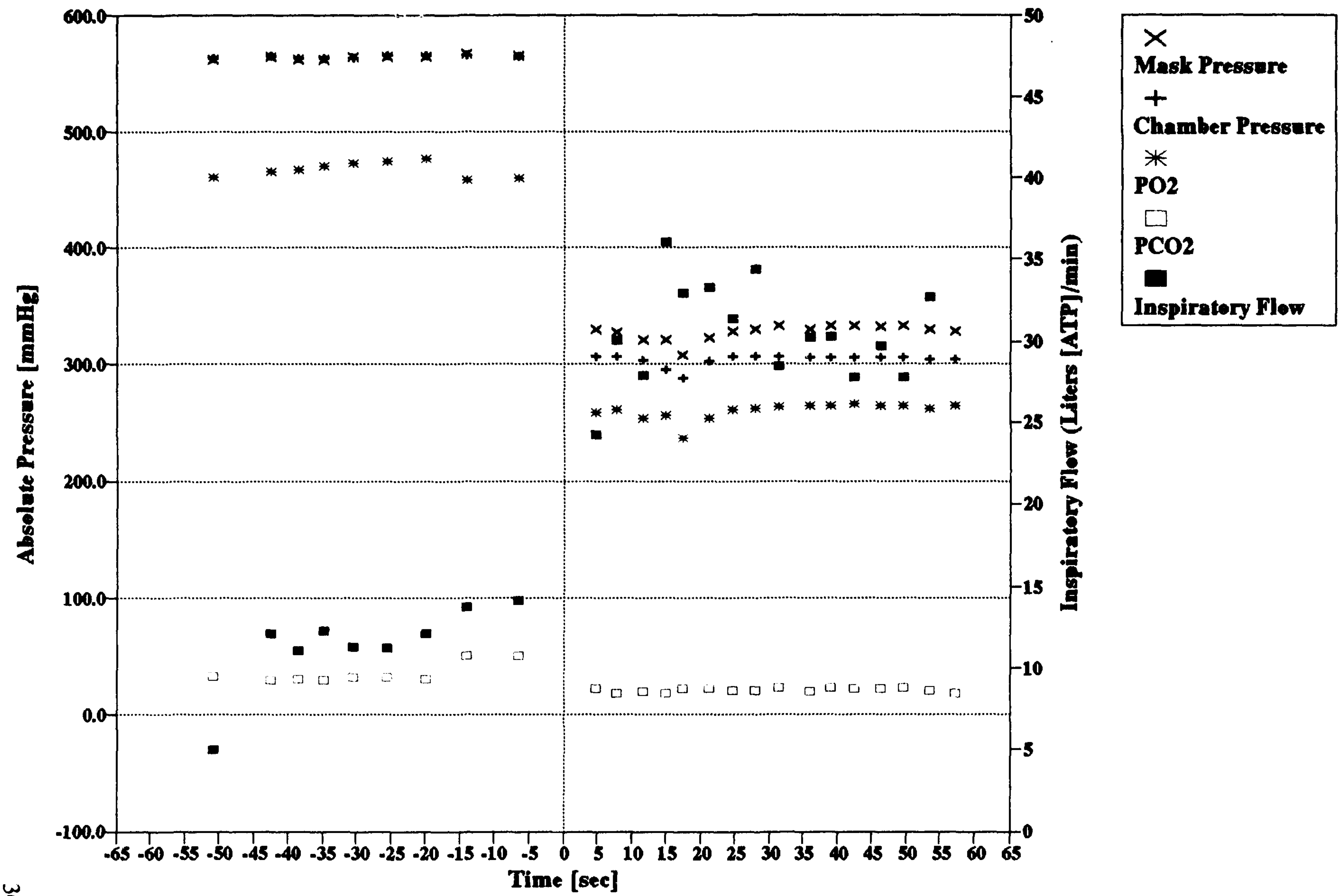
Subject: TED / 85% O2 Non-Dilution
8/20 kft Rapid Decompression



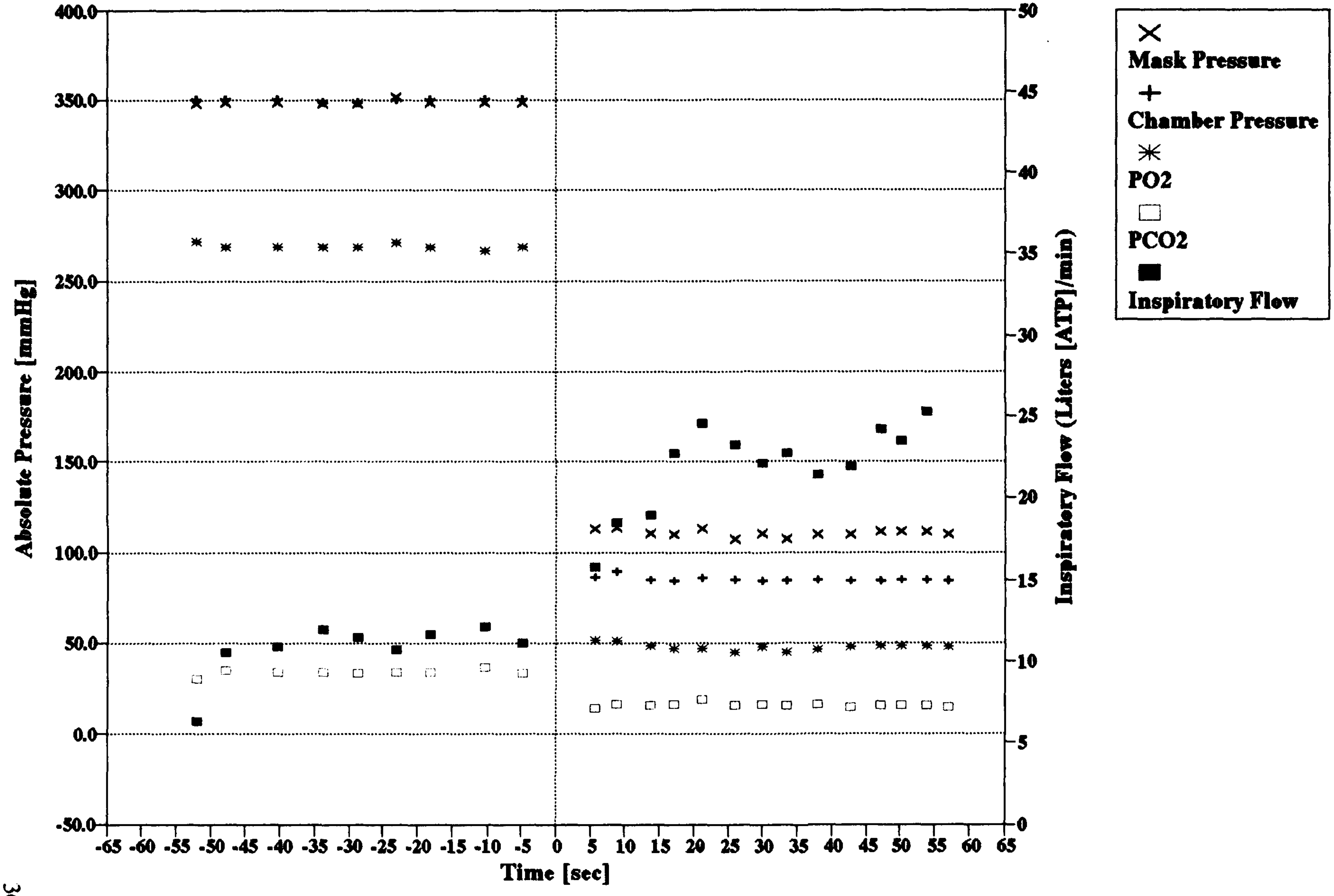
Subject: TED / 85% O2 Dilution
20/50 kft Rapid Decompression



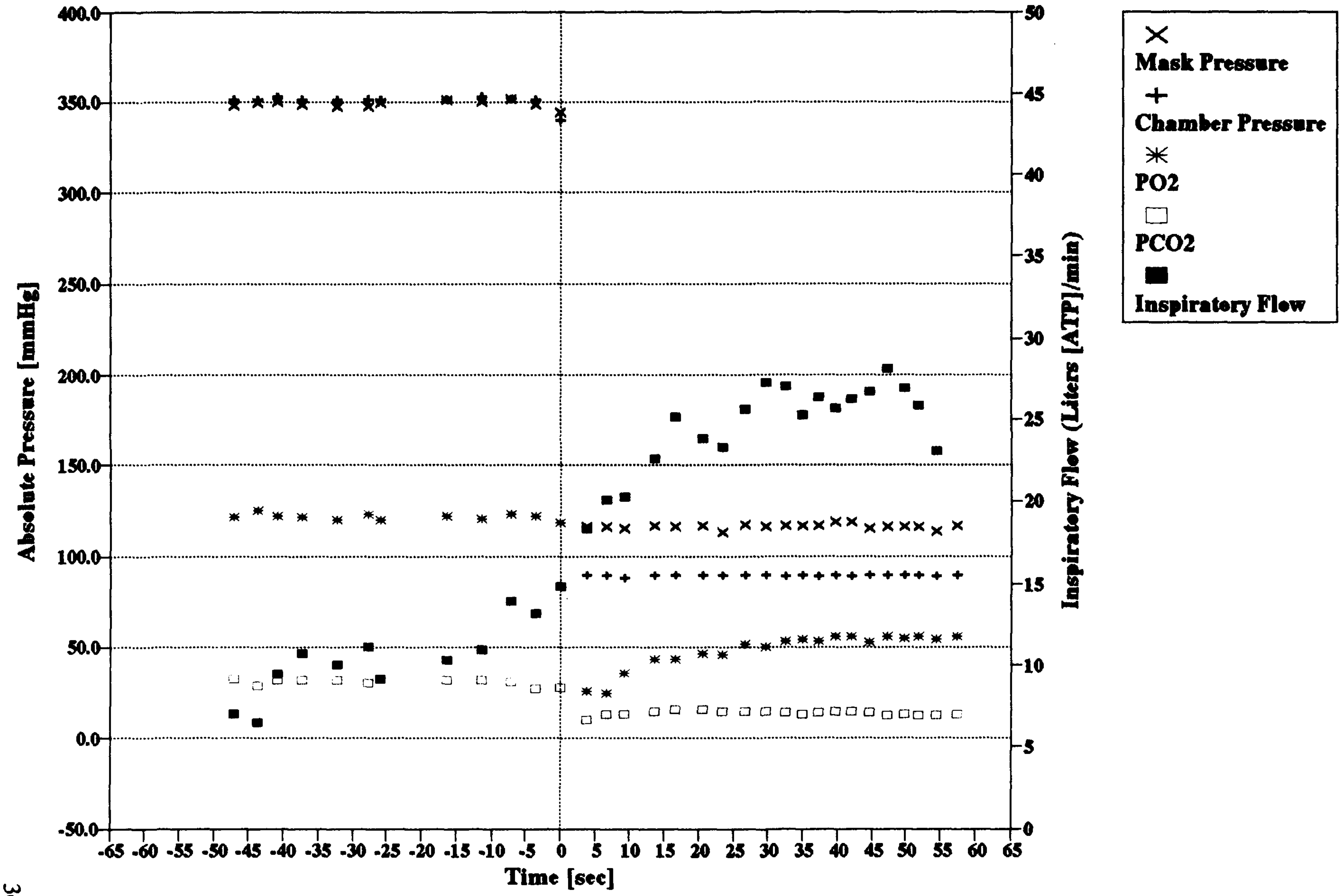
Subject: TED / 100% O2 Non-Dilution
8/20 kft Rapid Decompression



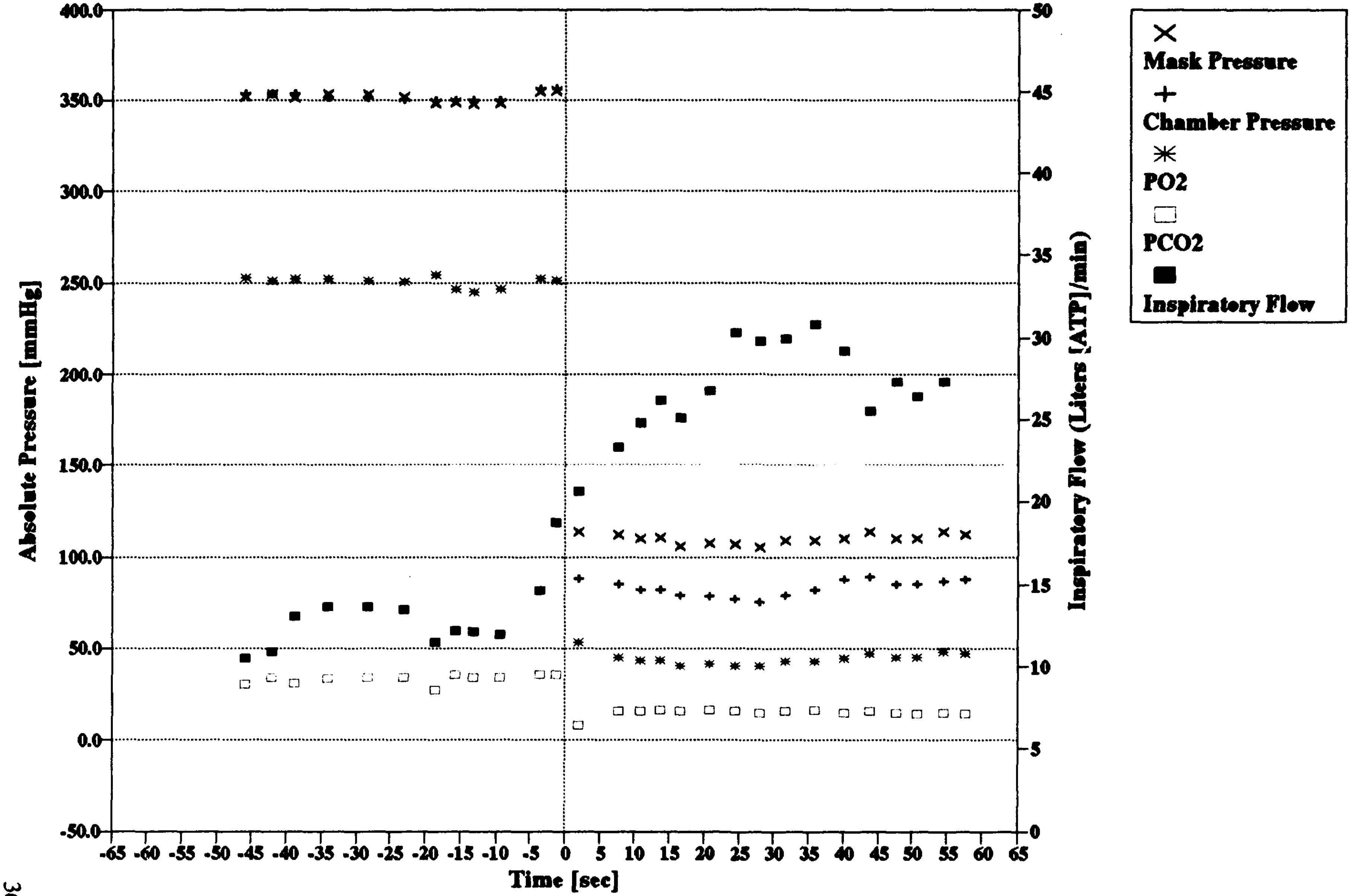
Subject: WRI / 100% O2 Non-Dilution
20/50 kft Rapid Decompression



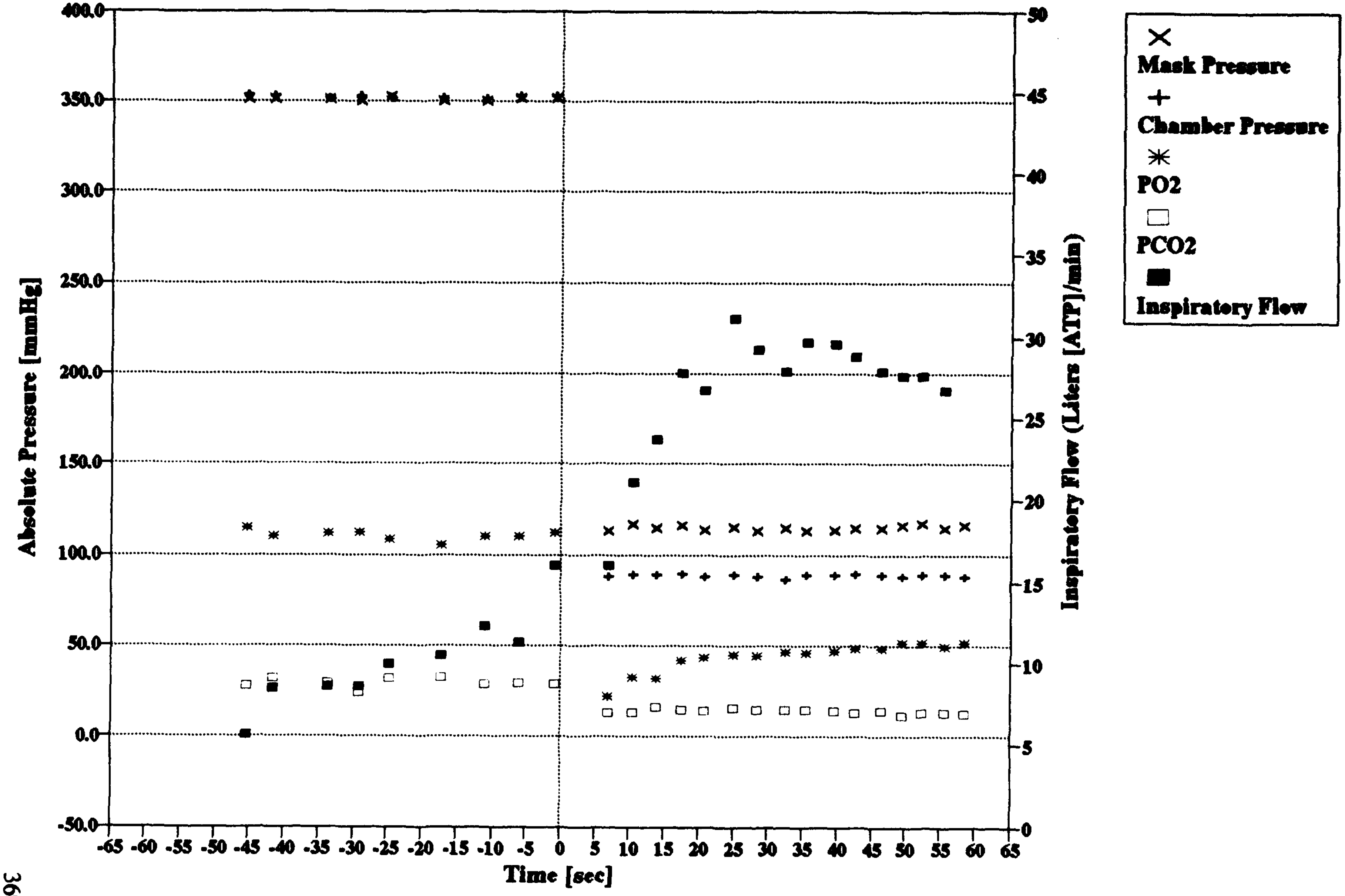
Subject: WRI / 100% O2 Dilution
20/50 kft Rapid Decompression

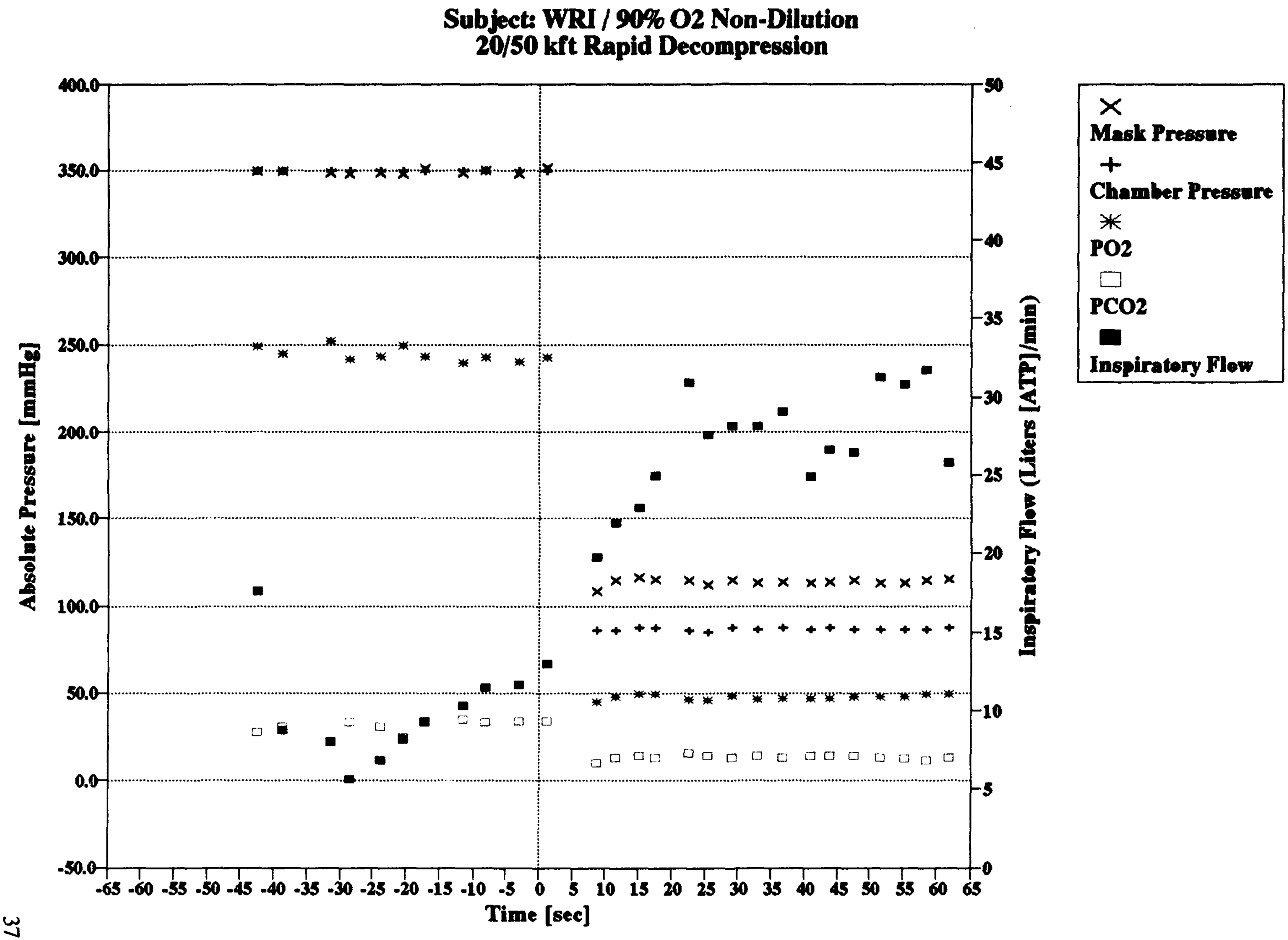


Subject: WRI / 93% O2 Non-Dilution
20/50 kft Rapid Decompression



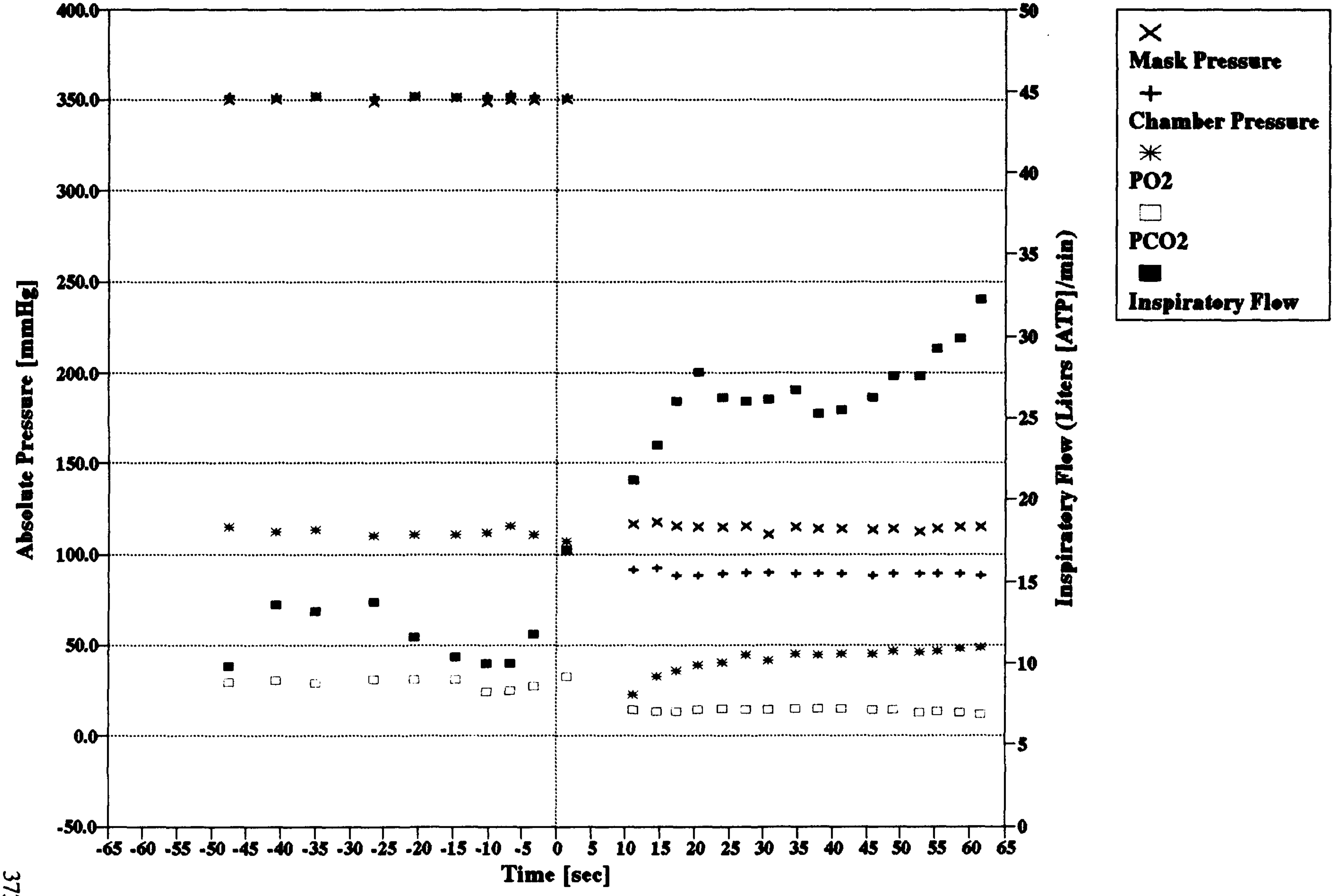
Subject: WRI / 93% O2 Dilution
20/50 kft Rapid Decompression



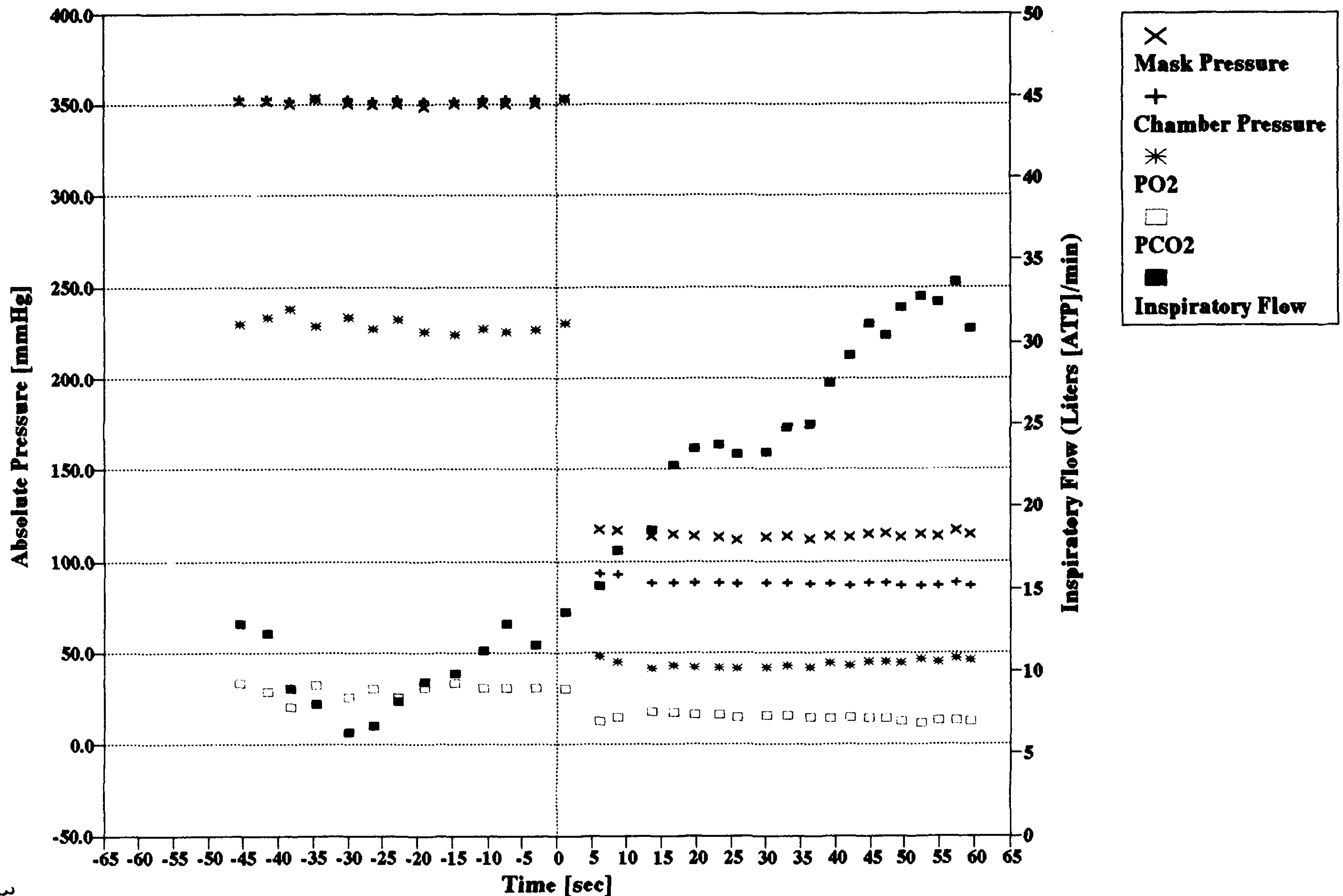


Subject: WRI / 90% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

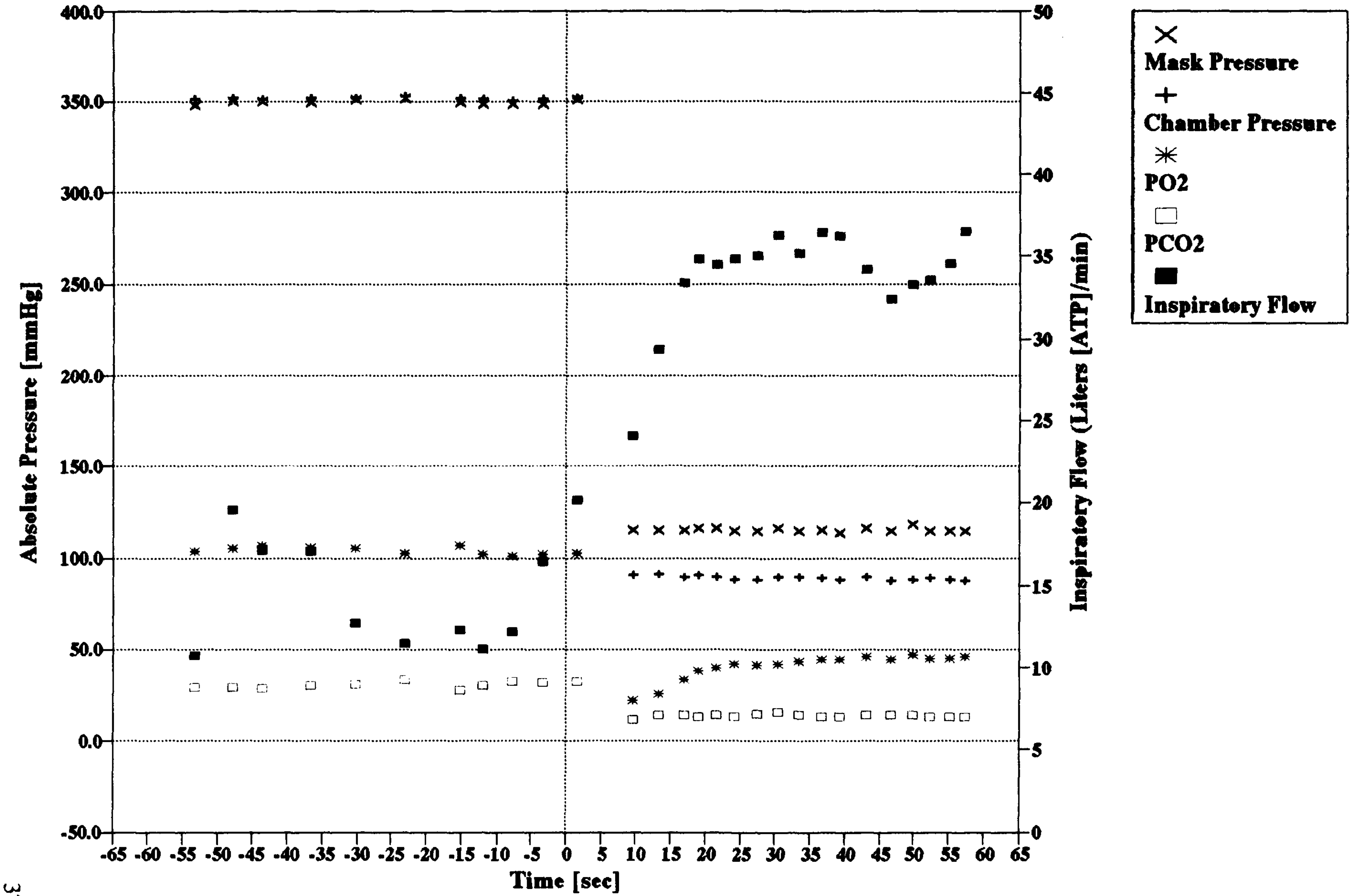
Subject: WRI / 90% O2 Dilution
20/50 kft Rapid Decompression

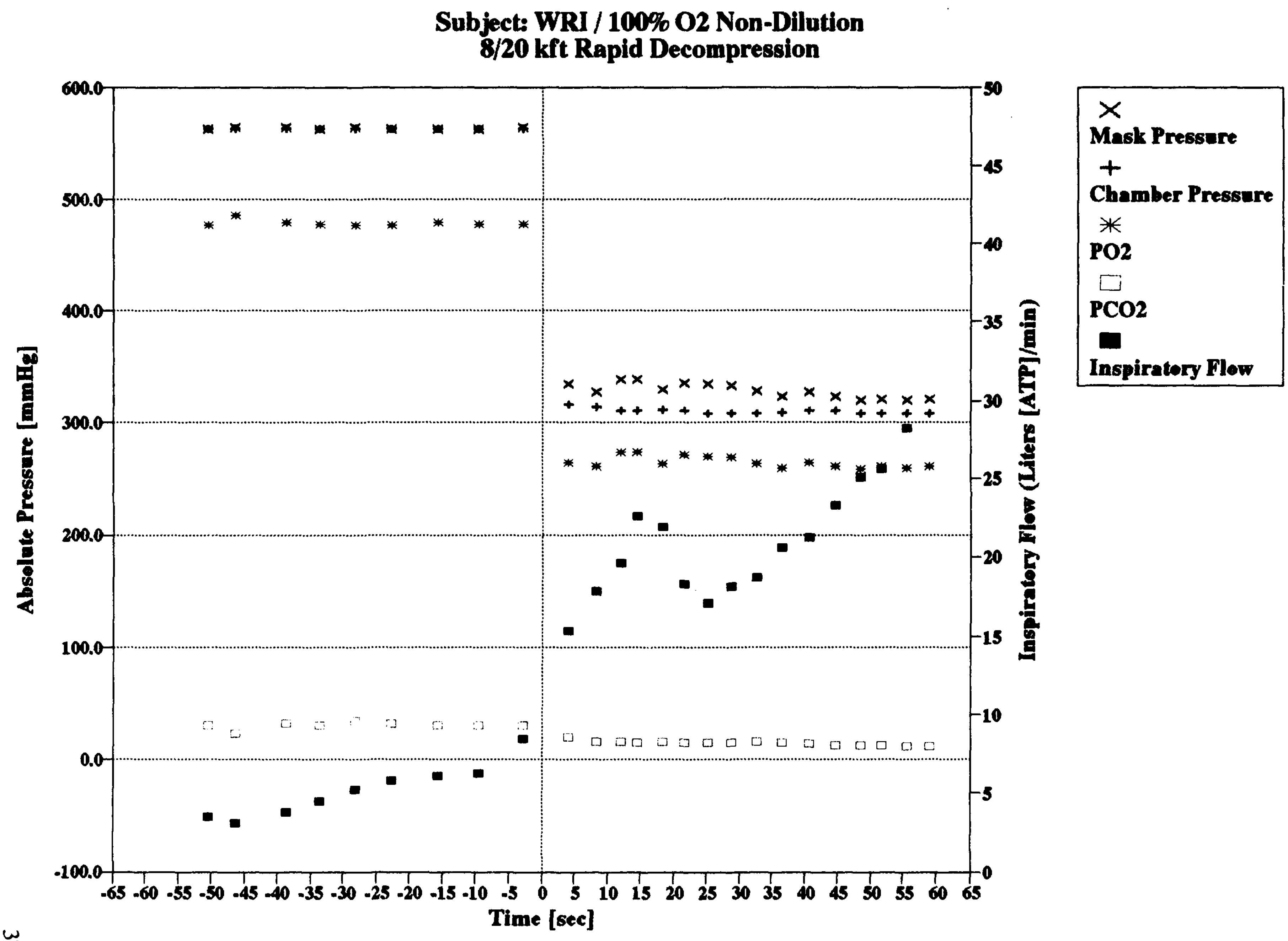


Subject: WRI / 85% O2 Non-Dilution
20/50 kft Rapid Decompression

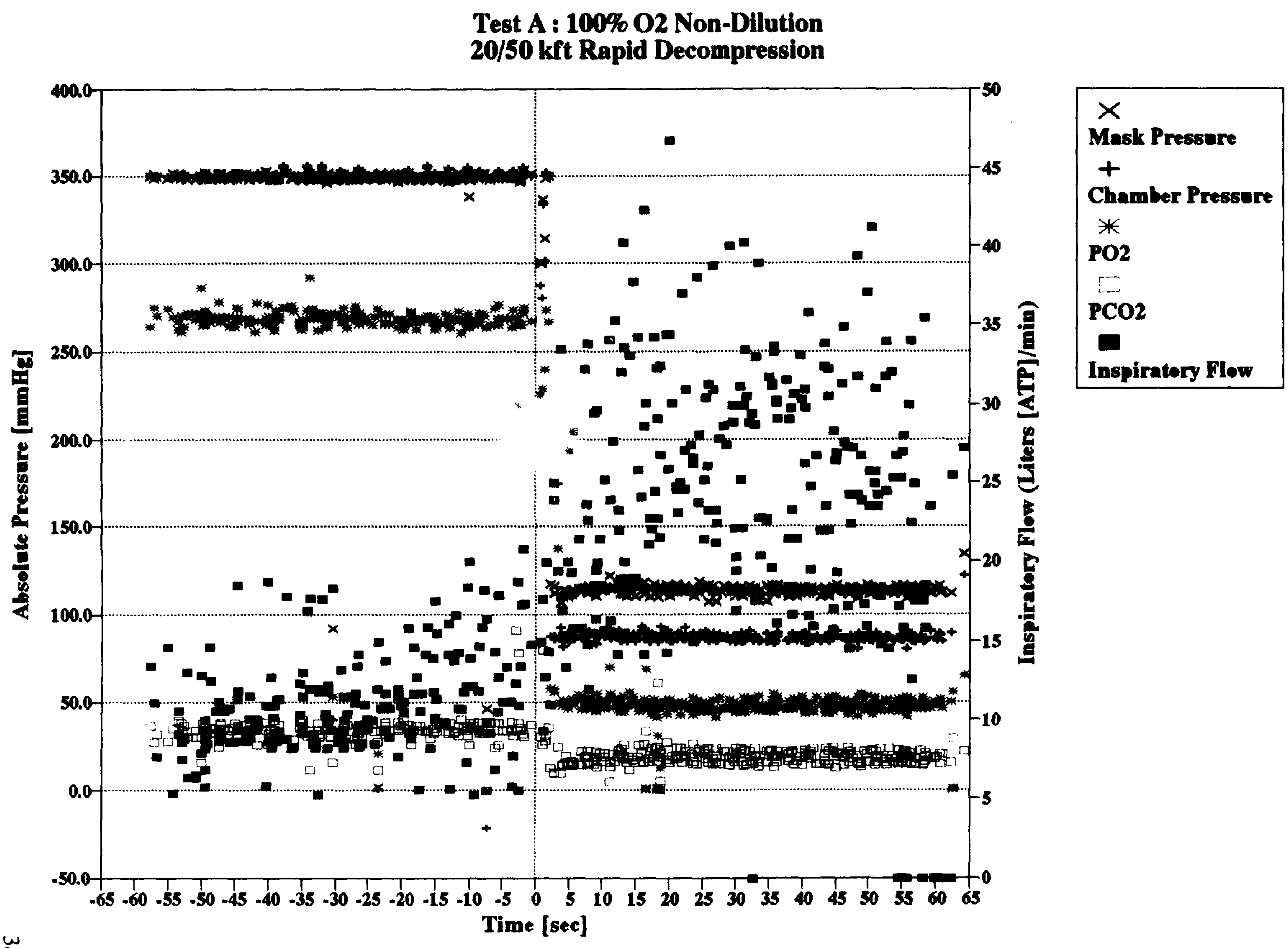


Subject: WRI / 85% O2 Dilution
20/50 kft Rapid Decompression

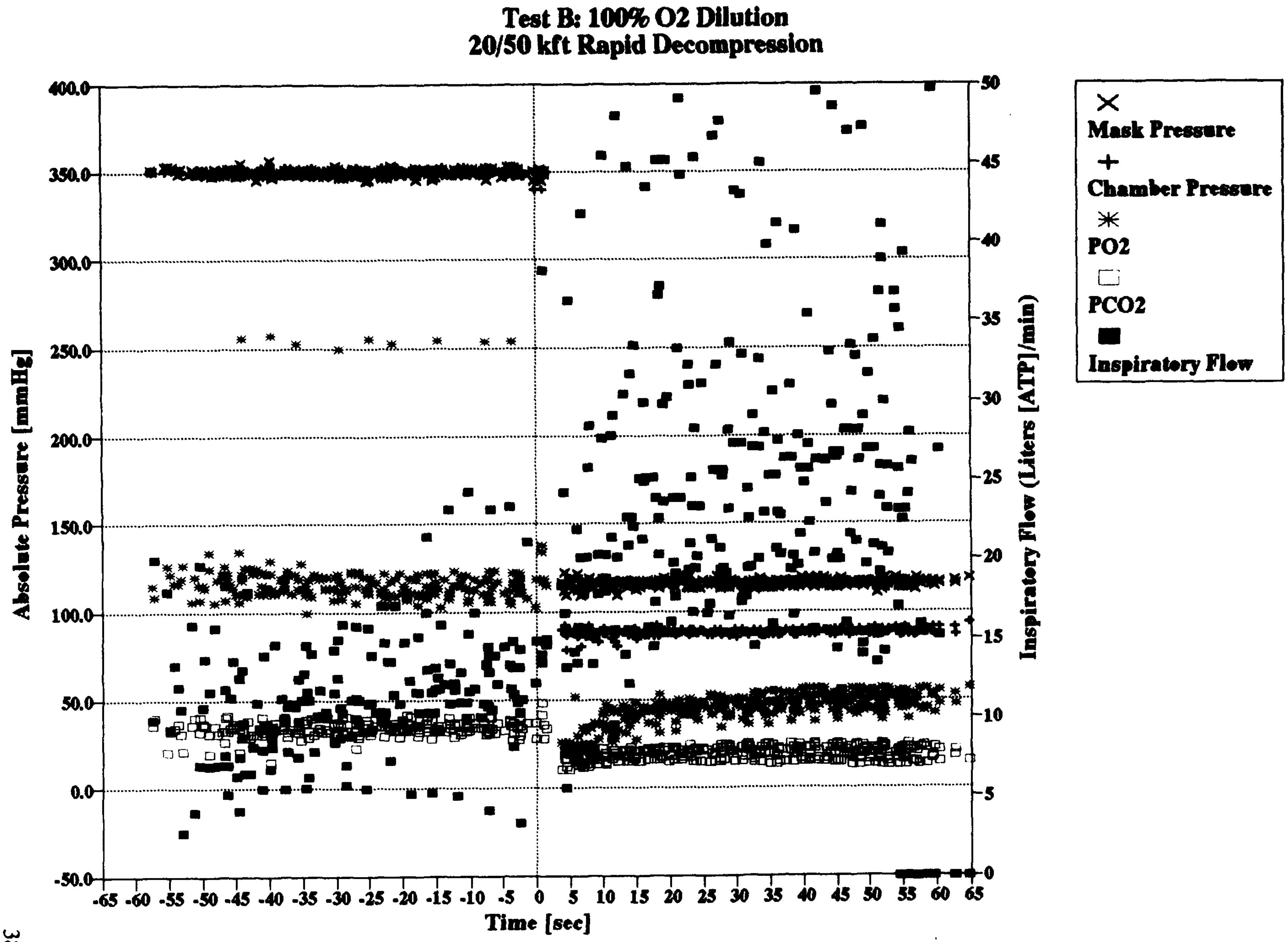




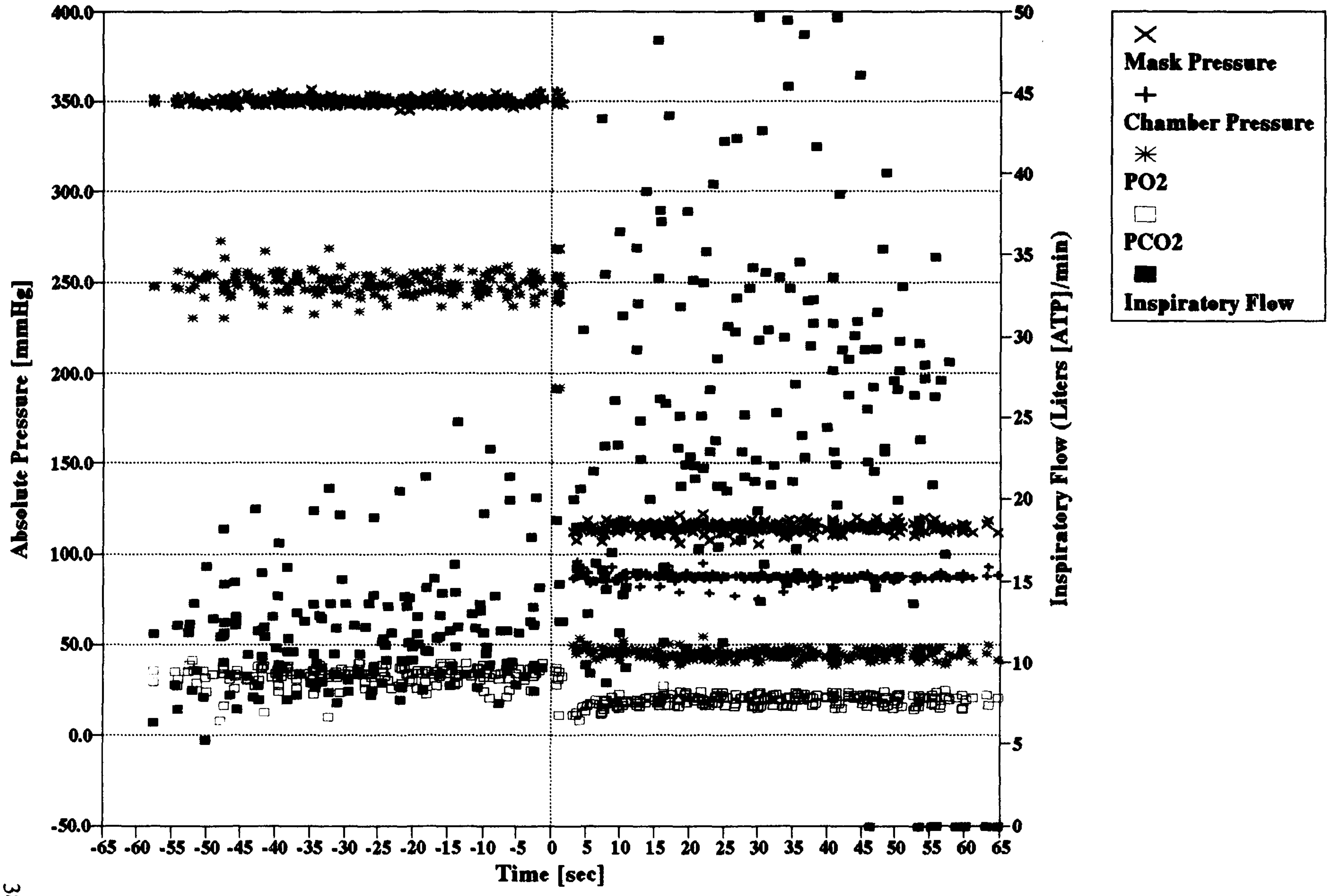
Subject: WRI / 100% O2 Non-Dilution
8/20 kft Rapid Decompression
Absolute Pressure [mmHg]
600.0
500.0
400.0
300.0
200.0
100.0
0.0
-100.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Time [sec]
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

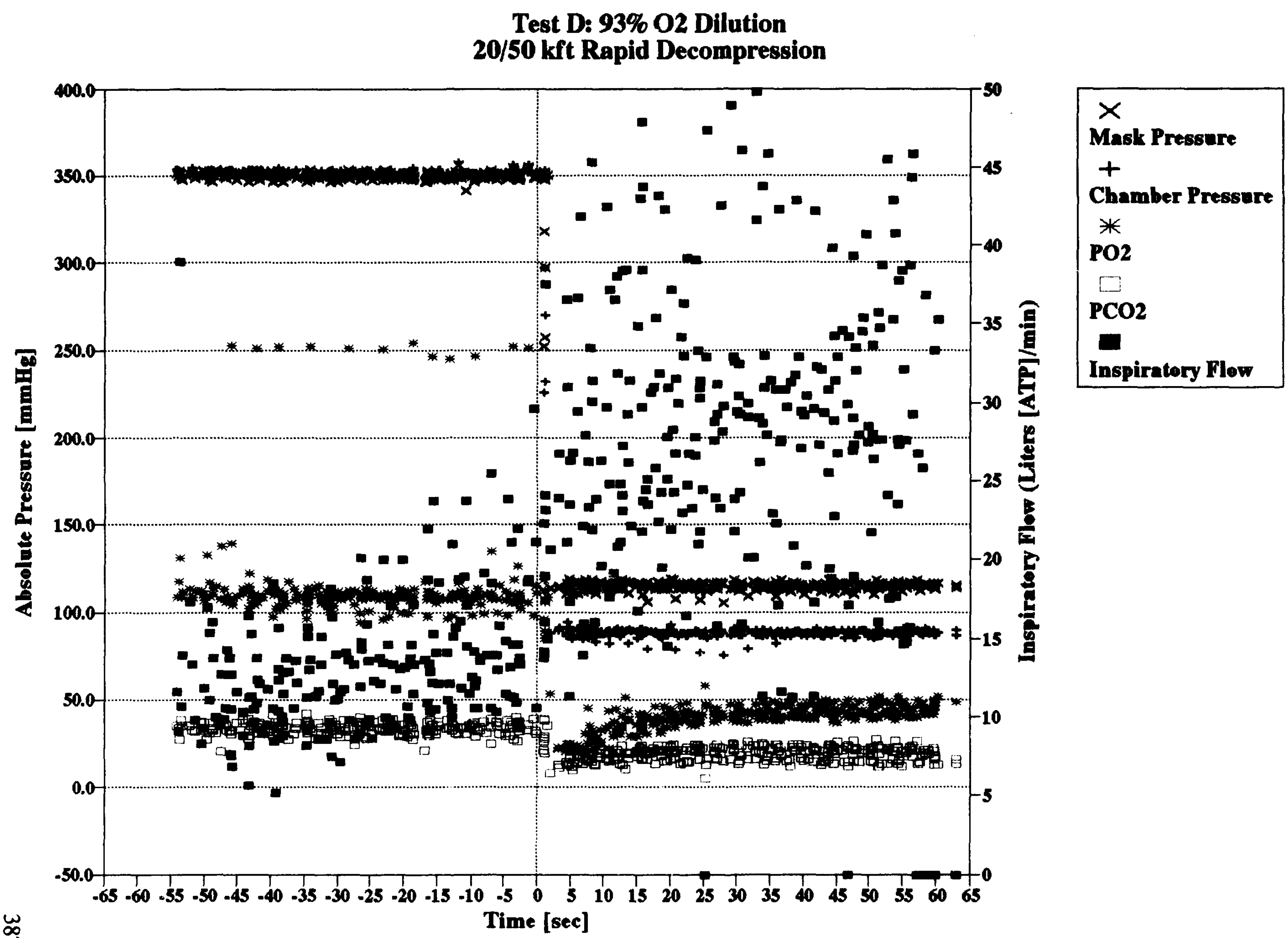


Test A : 100% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

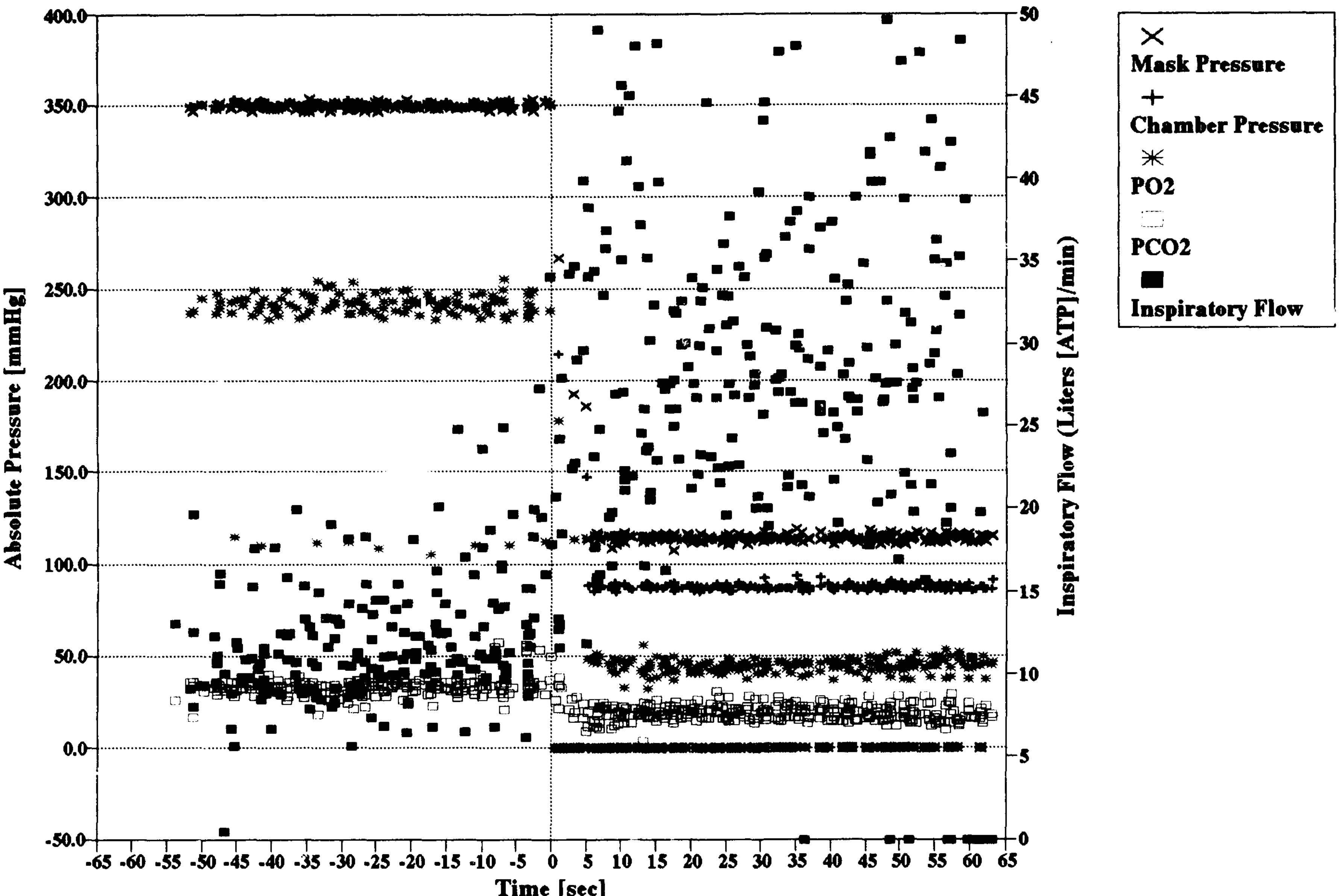

Test B: 100% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Inspiratory Flow (Liters [ATP]/min)
50
45
40
35
30
25
20
15
10
5
0
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

Test C: 93% O2 Non-Dilution
20/50 kft Rapid Decompression

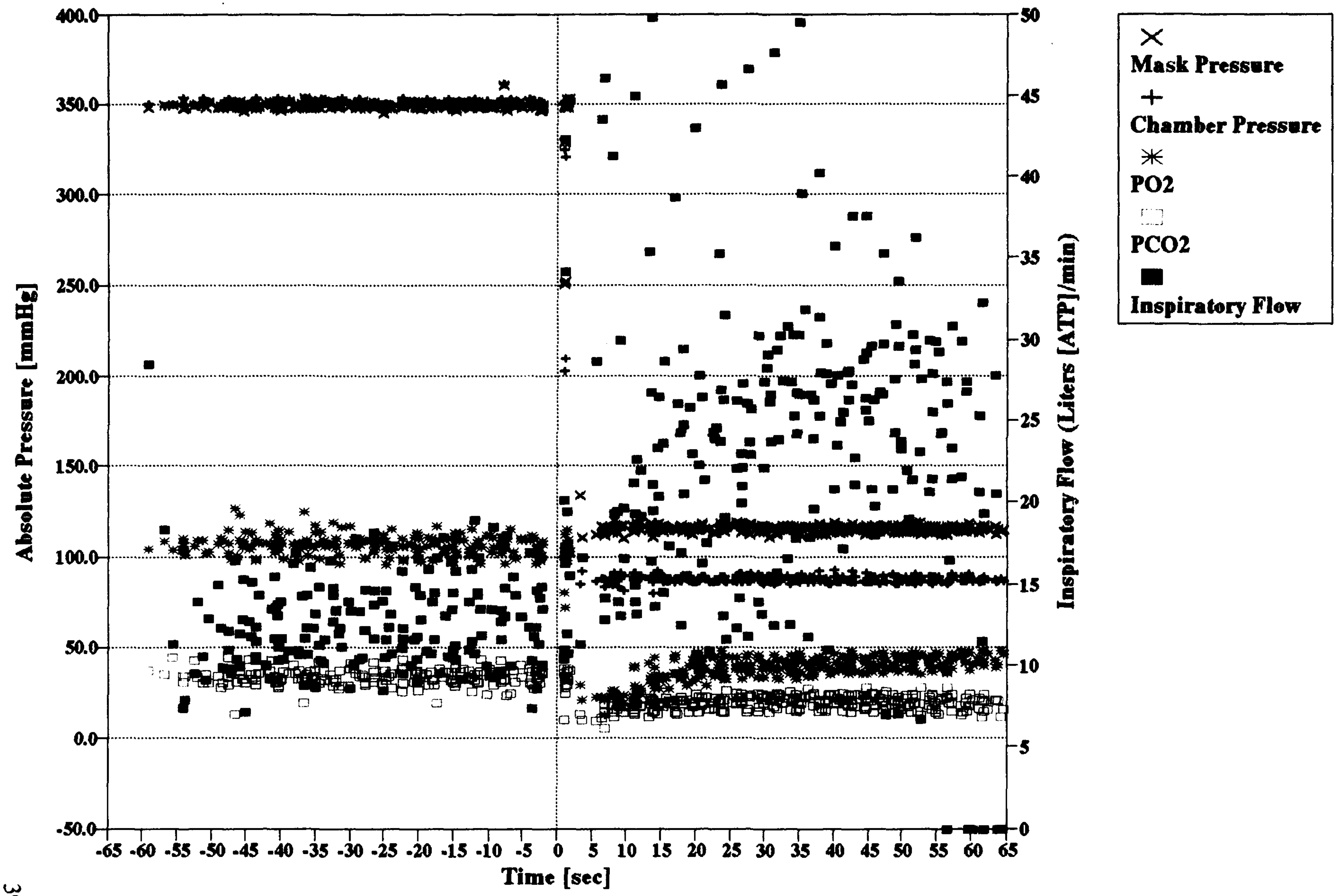



Test D: 93% O2 Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
Inspiratory Flow (Liters [ATP]/min)
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2
Inspiratory Flow

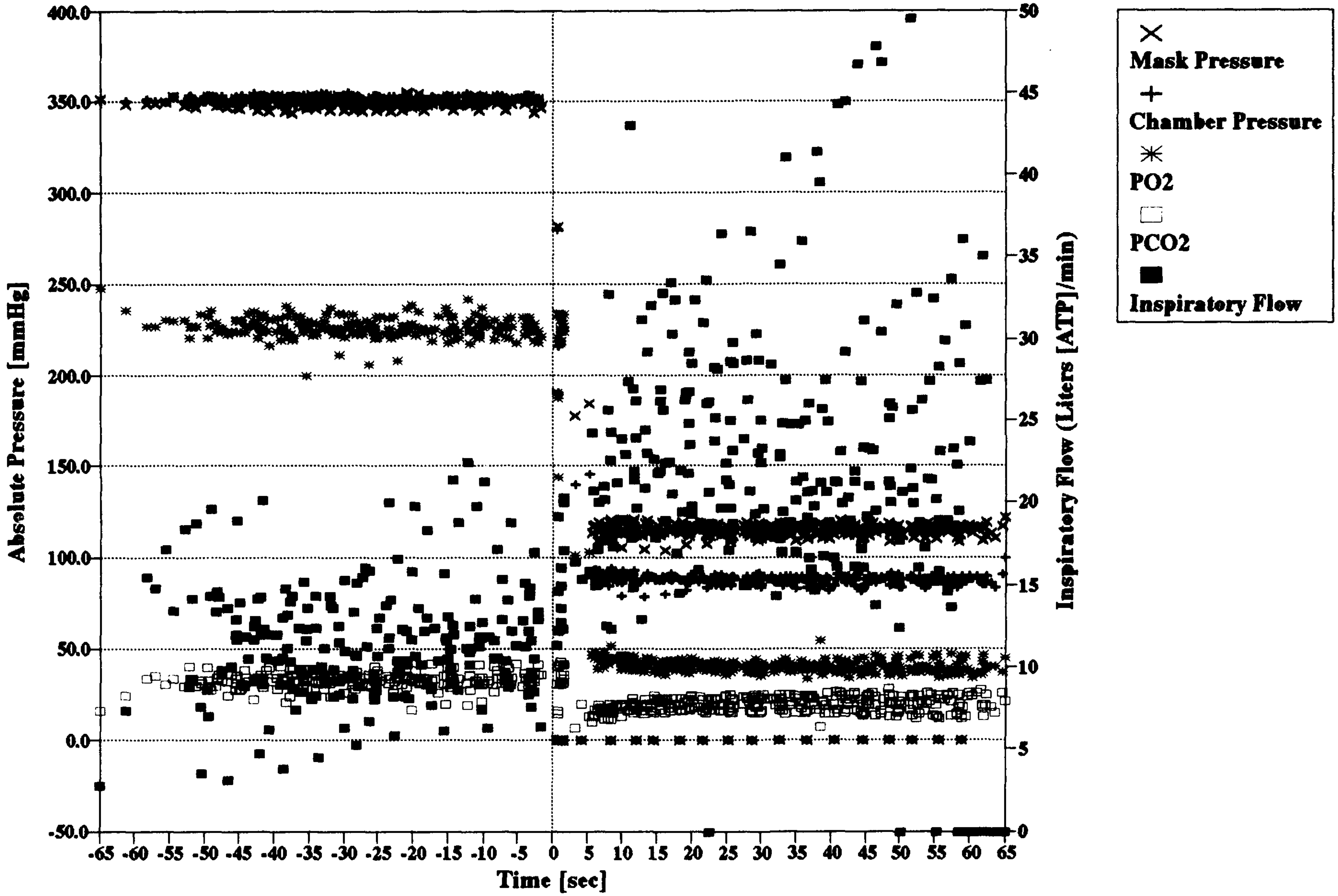
Test E: 90% O2 Non-Dilution
20/50 kft Rapid Decompression



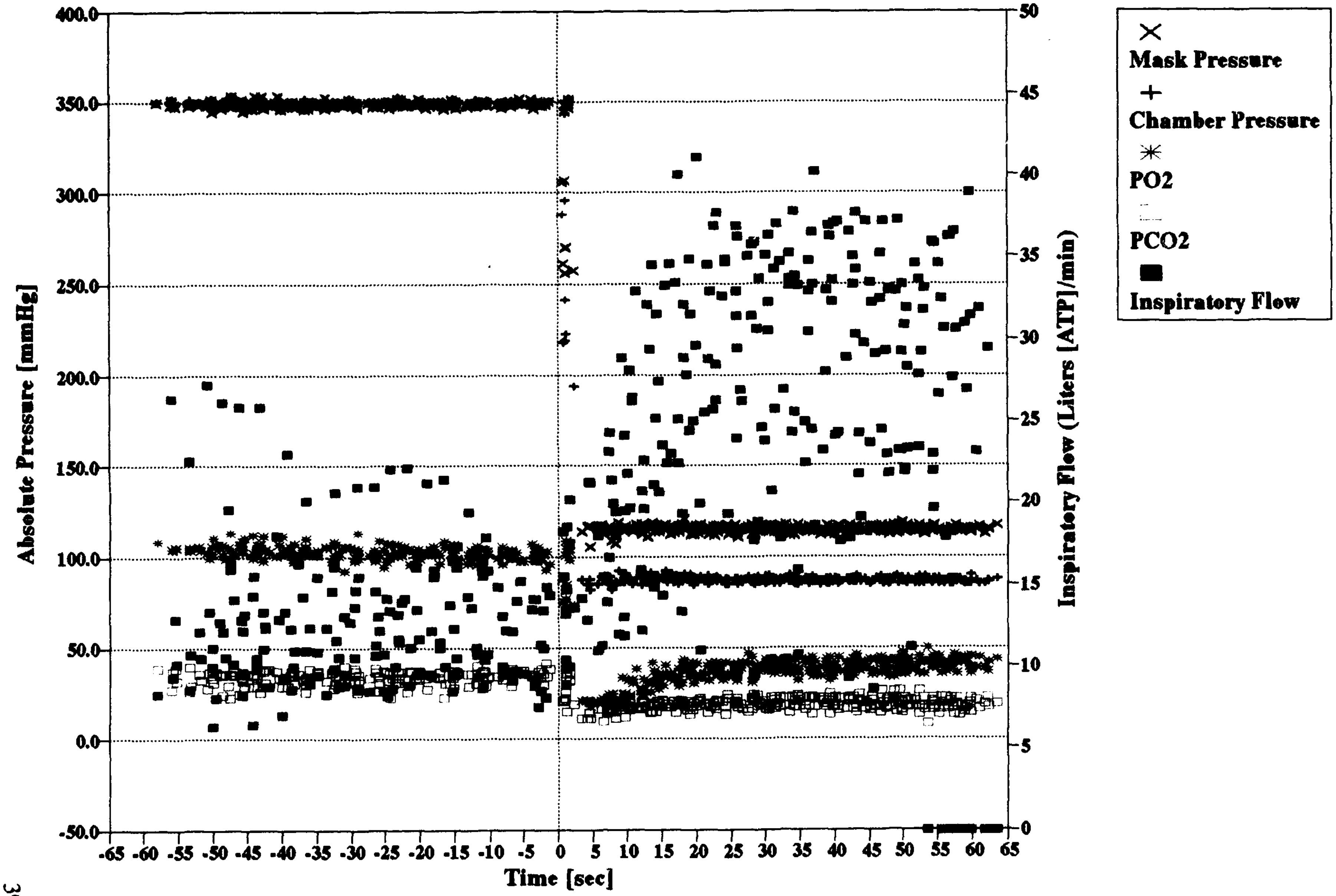
Test F: 90% O2 Dilution
20/50 kft Rapid Decompression



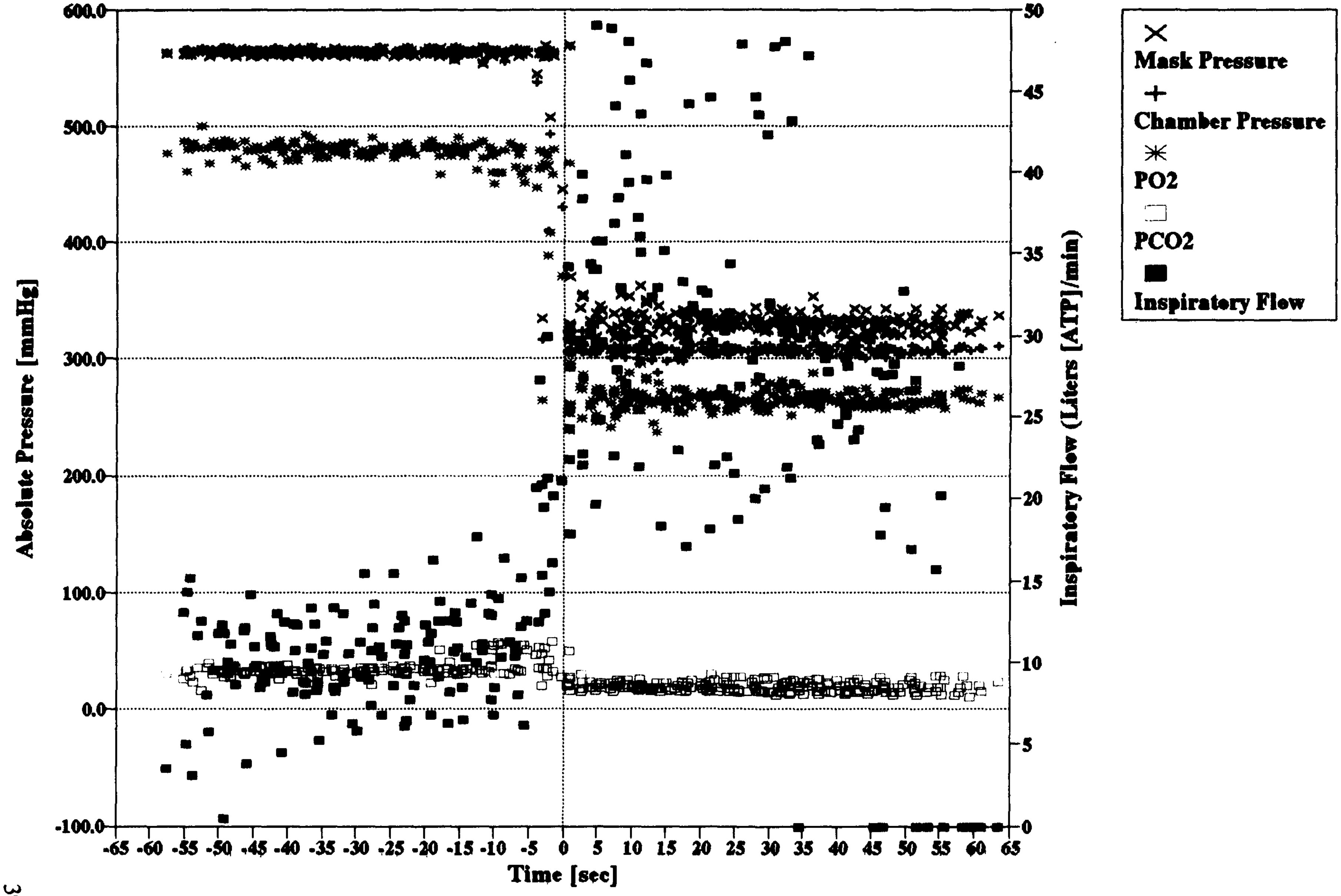
Test G: 85% O2 Non-Dilution
20/50 kft Rapid Decompression



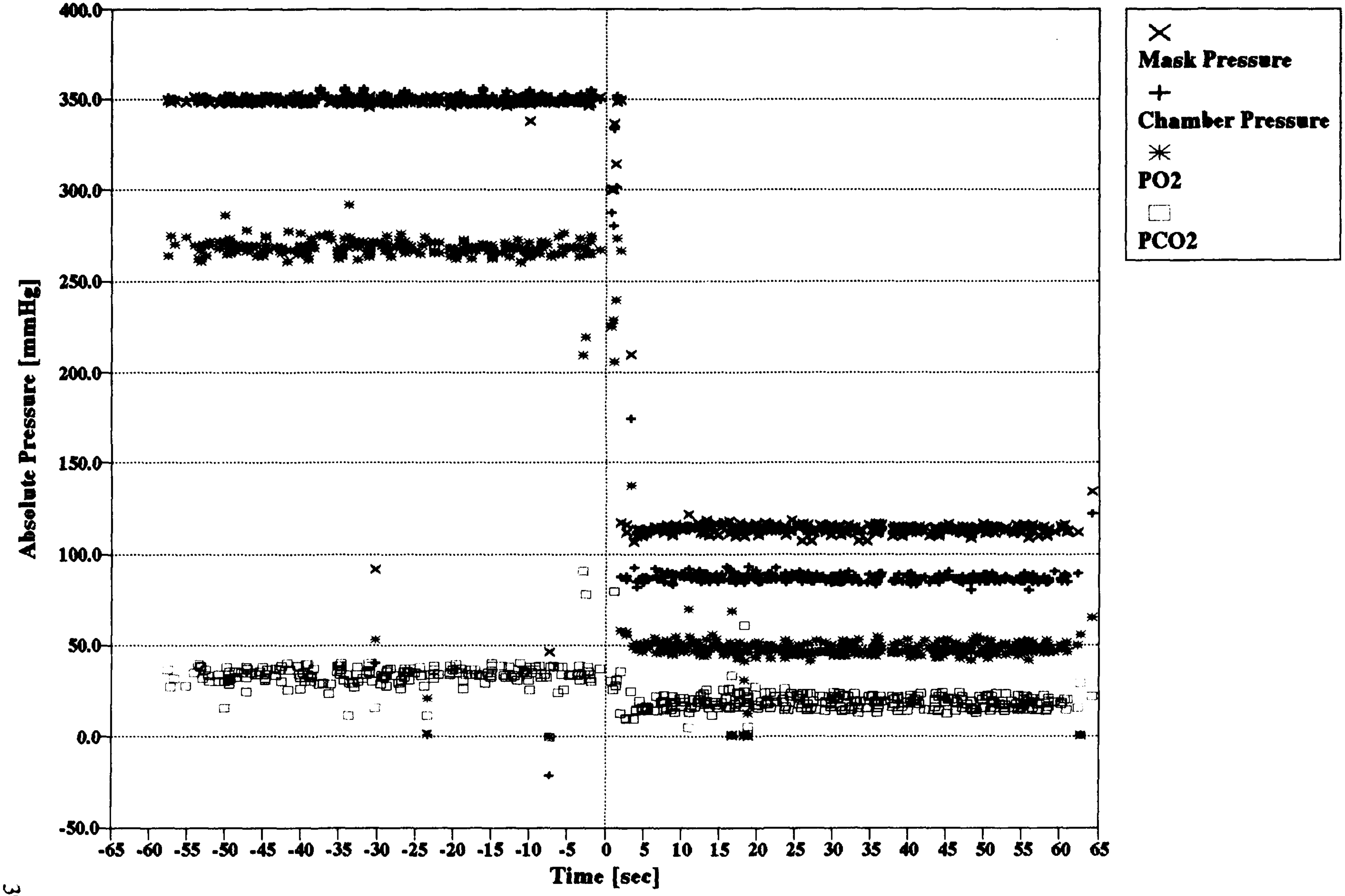
Test H: 85% O2 Dilution
20/50 kft Rapid Decompression



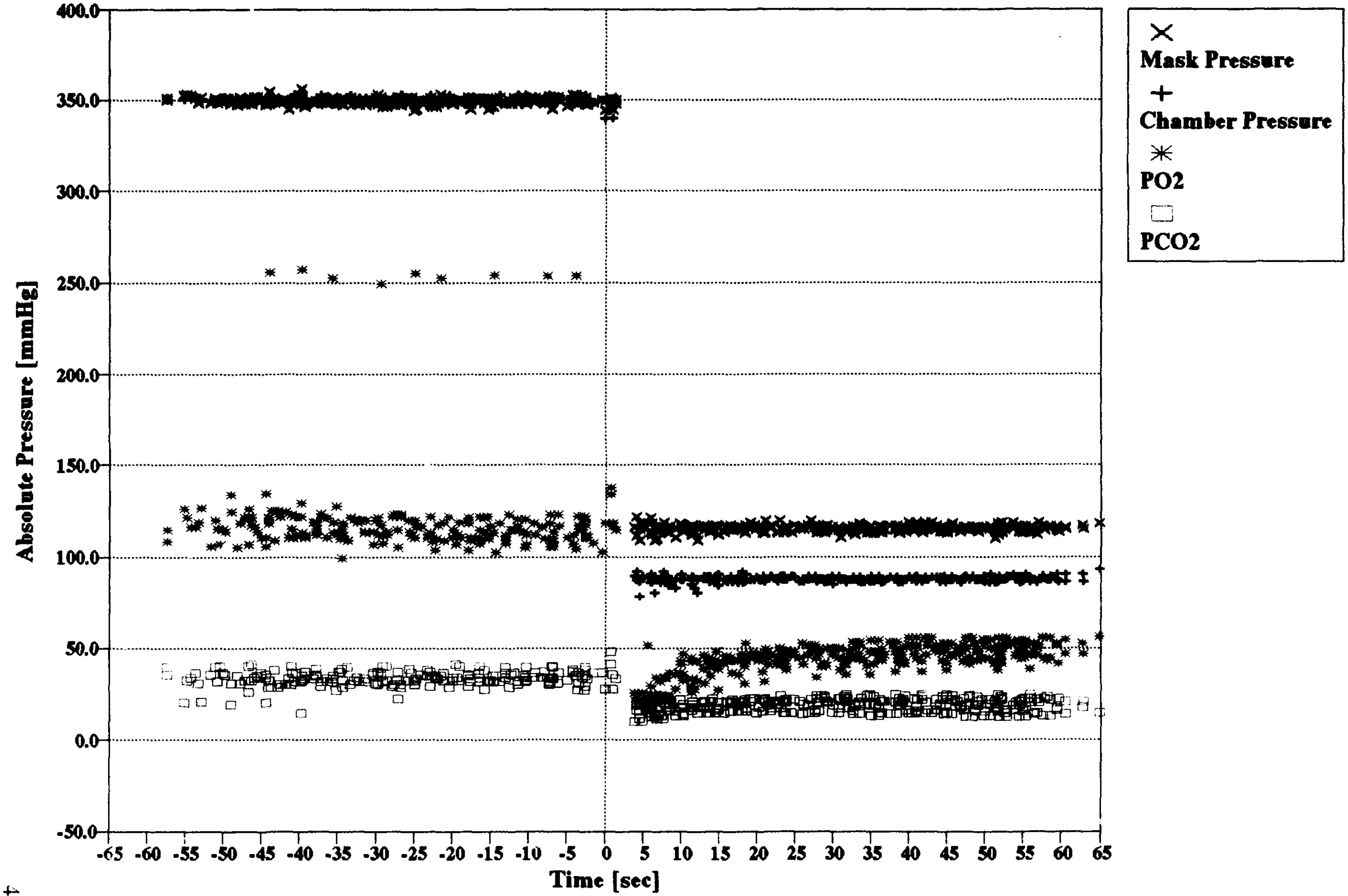
Test EONS2: 100% O2 Non-Dilution
8/20 kft Rapid Decompression

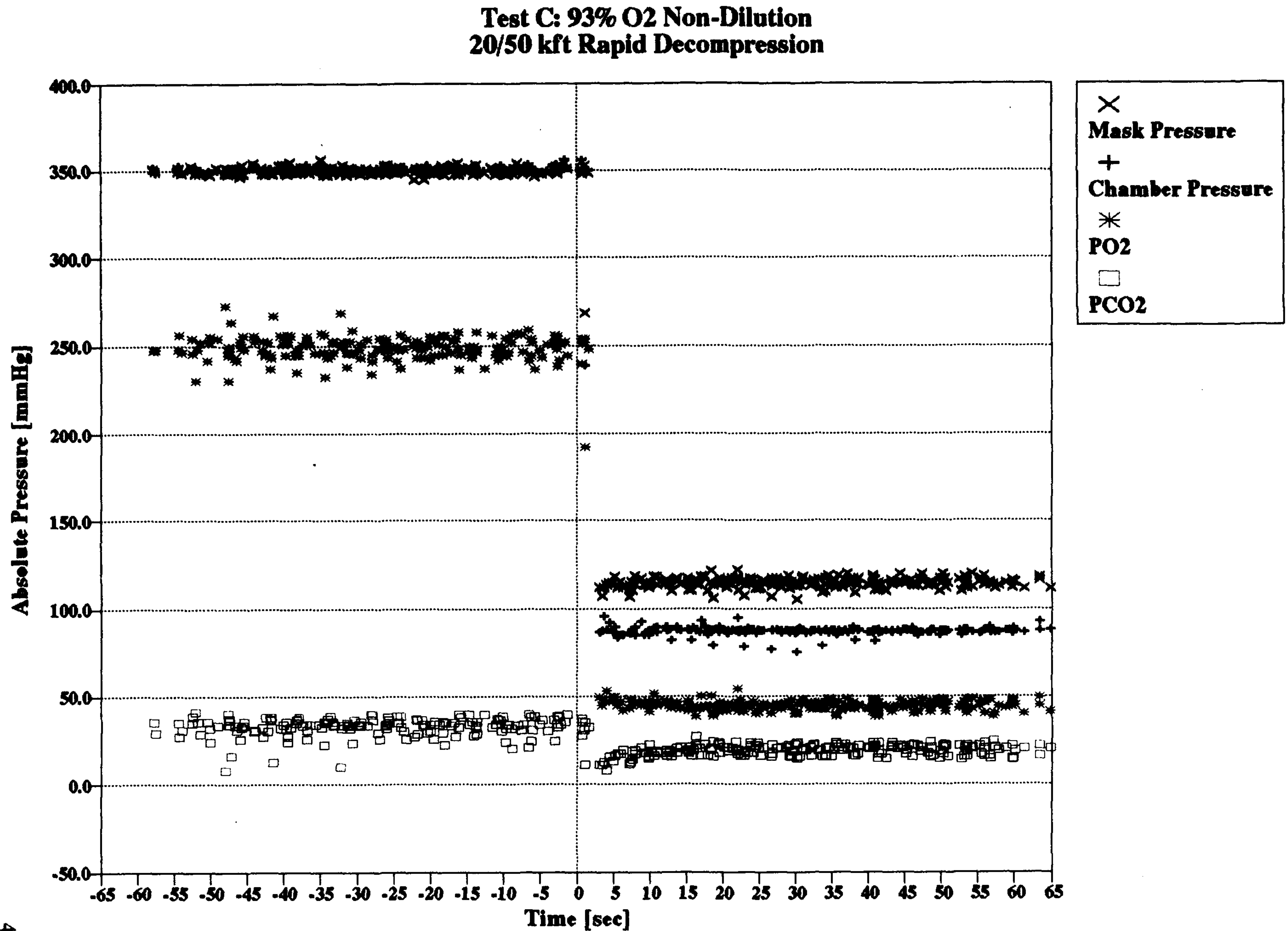


Test A : 100% O2 Non-Dilution
20/50 kft Rapid Decompression



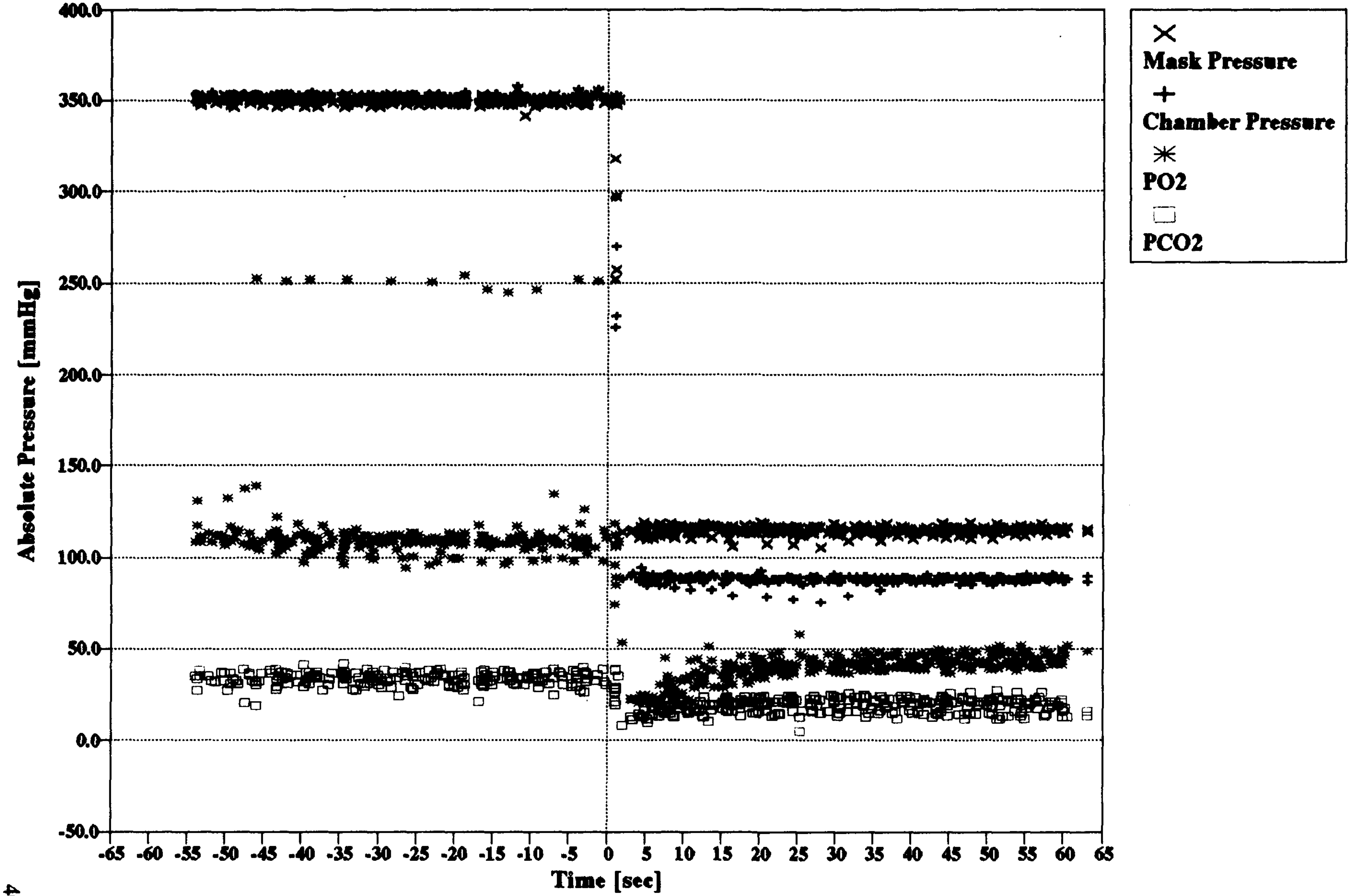
Test B: 100% O2 Dilution
20/50 kft Rapid Decompression



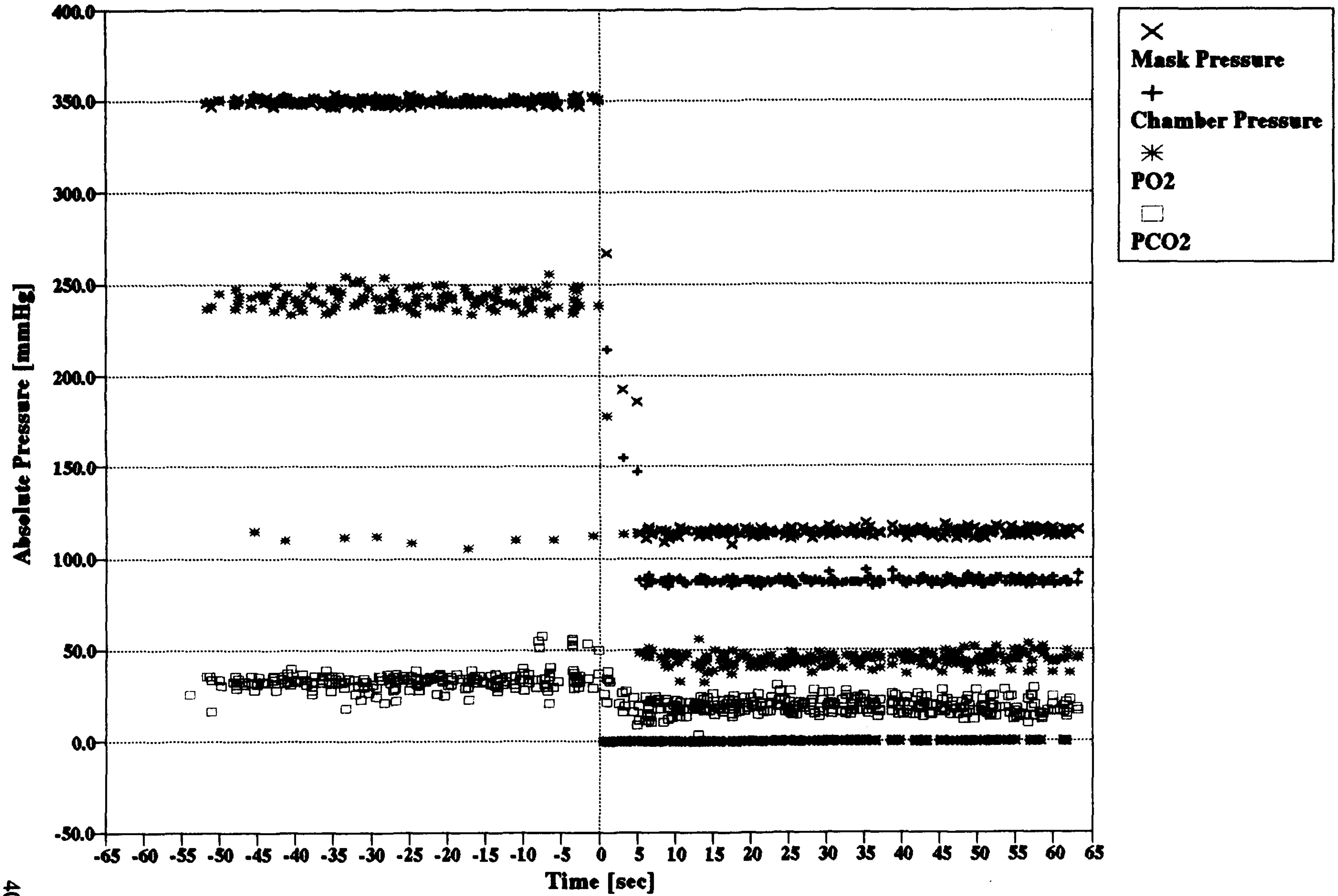


Test C: 93% O2 Non-Dilution
20/50 kft Rapid Decompression
Absolute Pressure [mmHg]
400.0
350.0
300.0
250.0
200.0
150.0
100.0
50.0
0.0
-50.0
-65 -60 -55 -50 -45 -40 -35 -30 -25 -20 -15 -10 -5 0 5 10 15 20 25 30 35 40 45 50 55 60 65
Time [sec]
Mask Pressure
Chamber Pressure
PO2
PCO2

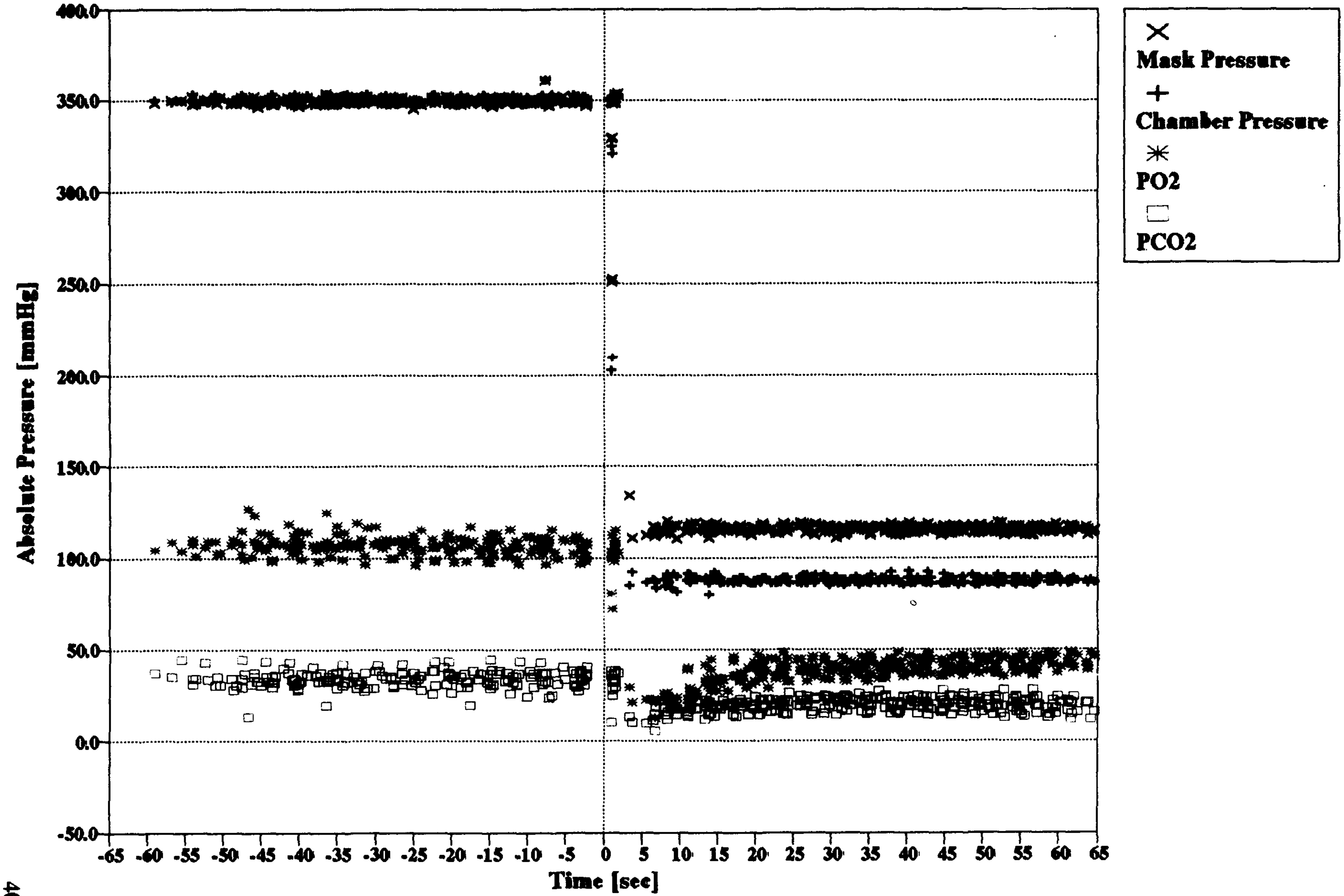
Test D: 93% O2 Dilution
20/50 kft Rapid Decompression



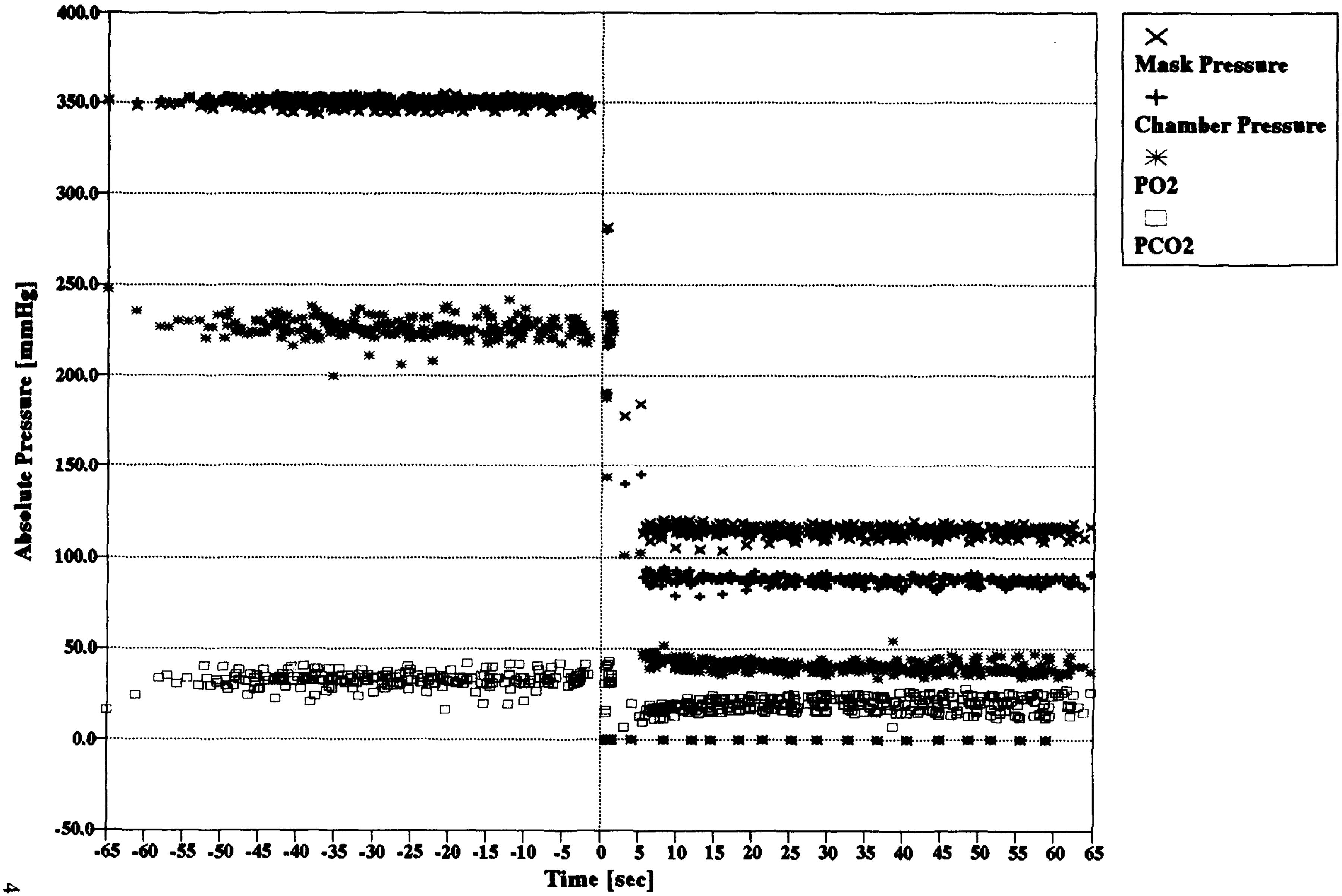
Test E: 90% O2 Non-Dilution
20/50 kft Rapid Decompression



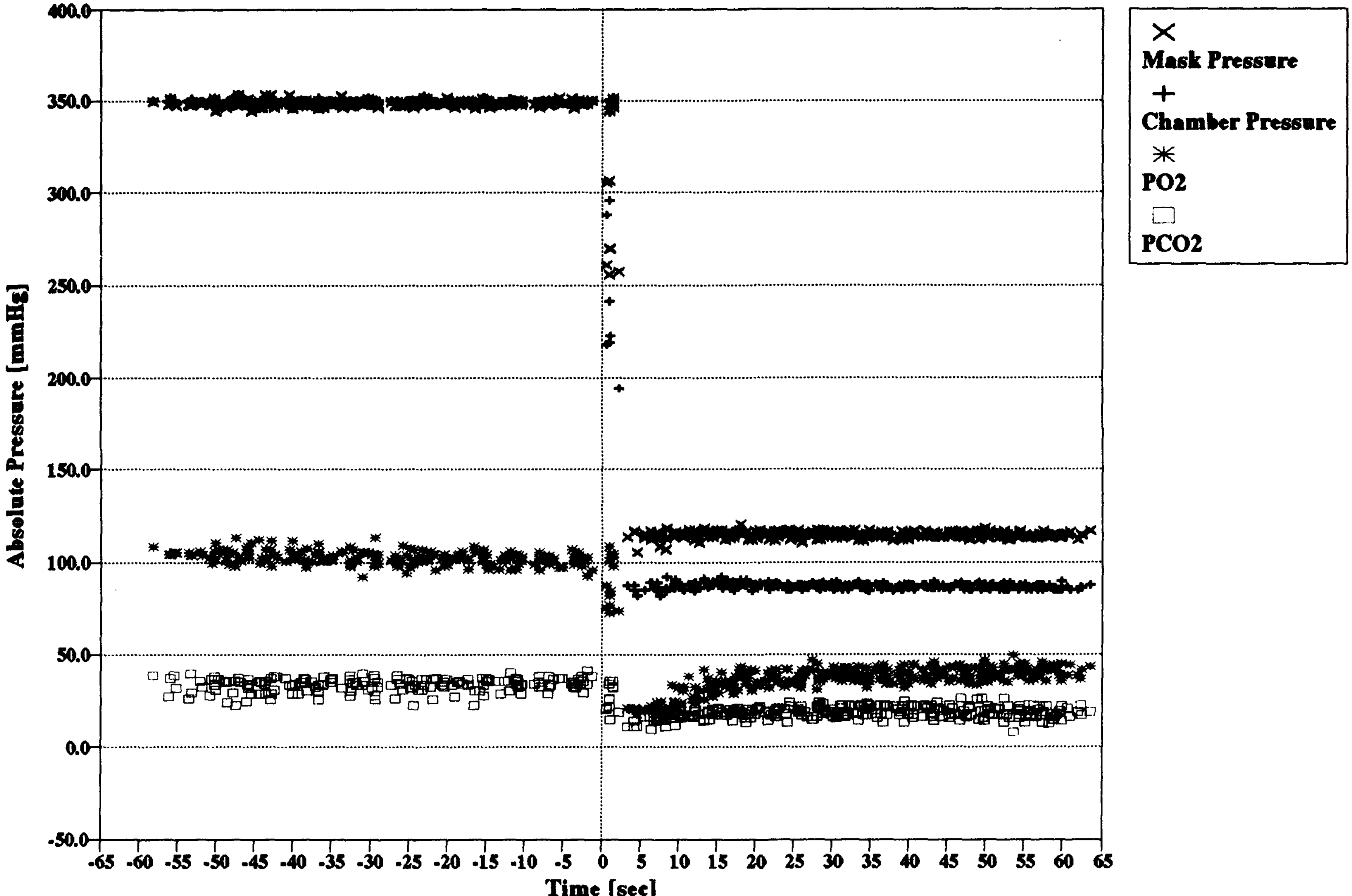
Test F: 90% O2 Dilution
20/50 kft Rapid Decompression



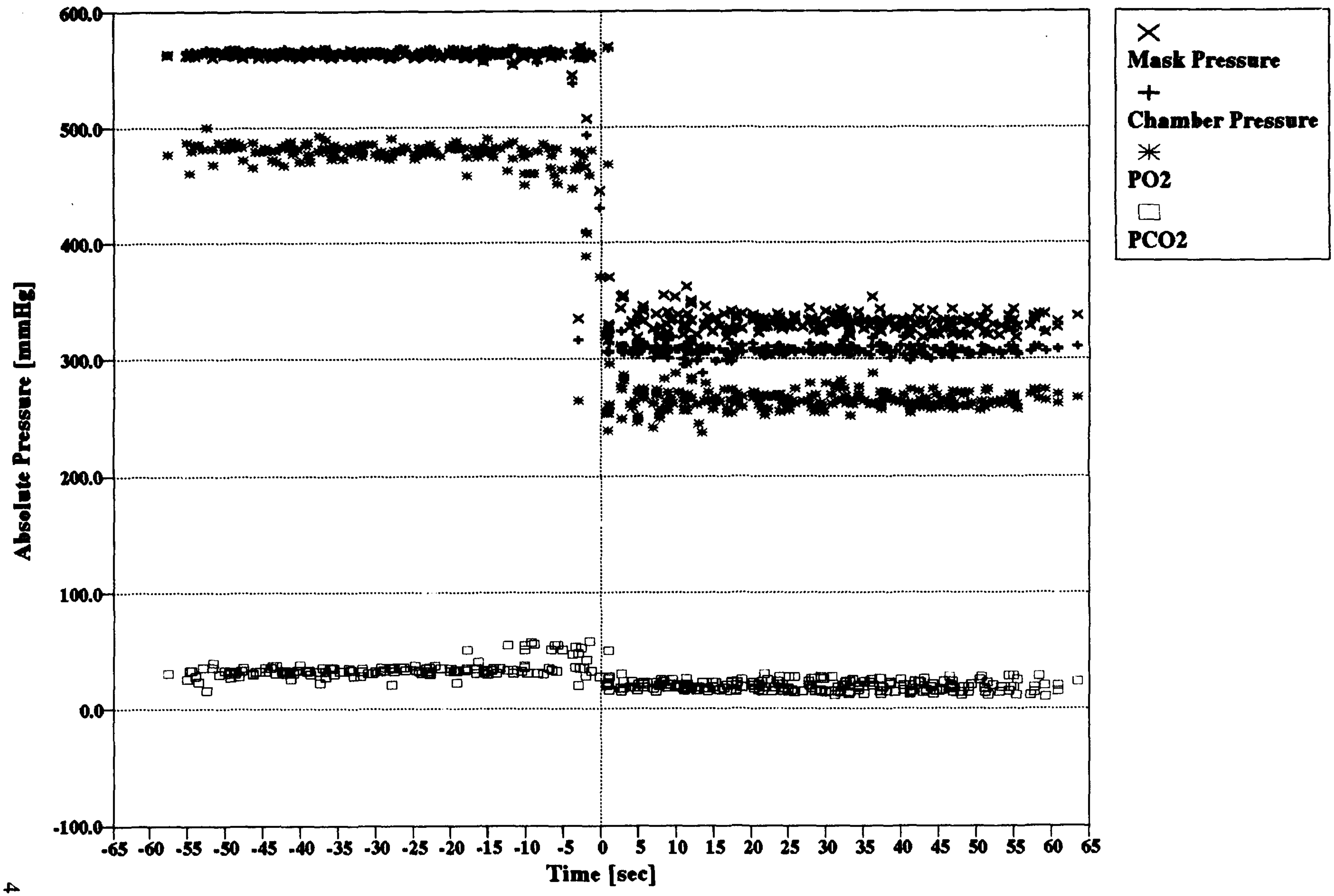
Test G: 85% O2 Non-Dilution
20/50 kft Rapid Decompression



Test H: 85% O2 Dilution
20/50 kft Rapid Decompression



Test EONS2: 100% O2 Non-Dilution
8/20 kft Rapid Decompression



APPENDIX B

Mask/Regulator Model Computer Program

Appendix B
Mask/Regulator Model Computer Program

PROGRAM MASKMODL

```
C...
C...  PROGRAM MASKMODL CALLS: (1) SUBROUTINE INITIAL TO DEFINE THE ODE
C...  INITIAL CONDITIONS, (2) SUBROUTINE RKF45 TO INTEGRATE THE ODES,
C...  AND (3) SUBROUTINE PRINT TO PRINT THE SOLUTION.
C...
C...  THE FOLLOWING CODING IS FOR 500 ODES.  IF MORE ODES ARE TO BE INTE-
C...  GRATED, ALL OF THE 500'S SHOULD BE CHANGED TO THE REQUIRED NUMBER
      IMPLICIT DOUBLE PRECISION (A-H), DOUBLE PRECISION (O-Z)
      INTEGER NI, NO, NEQN, NSTOP, NORUN
      COMMON/T/          T, XT,    NSTOP,     NORUN
     1      /Y/    Y(500)
     2      /F/    F(500)
C...
C...  THE NUMBER OF DIFFERENTIAL EQUATIONS IS IN COMMON/N/ FOR USE IN
C...  SUBROUTINE FCN
      COMMON/N/       NEQN
C...
C...  COMMON AREA TO PROVIDE THE INPUT/OUTPUT UNIT NUMBERS TO OTHER
C...  SUBROUTINES
      COMMON/IO/       NI,        NO
C...
C...  ABSOLUTE DIMENSIONING OF THE ARRAYS REQUIRED BY RKF45
      DOUBLE PRECISION YV(500), WORK(11000)
      INTEGER IWORK(5)
C...
C...  EXTERNAL THE DERIVATIVE ROUTINE CALLED BY RKF45
      EXTERNAL FCN
C...
C...  ARRAY FOR THE TITLE (FIRST LINE OF DATA), CHARACTERS  END OF RUNS
      CHARACTER TITLE(20)*4, ENDRUN(3)*4
C...
C...  DEFINE THE CHARACTERS  END OF RUNS
      DATA ENDRUN/'END ','OF R','UNS '/
C...
C...  DEFINE THE INPUT/OUTPUT UNIT NUMBERS
      NI=5
      NO=6
C...
C...  OPEN INPUT AND OUTPUT FILES
      OPEN(NI,FILE='MASKDAT.DAT')
      OPEN(NO,FILE='OUTPUT.PRN',BLOCKSIZE=2048)
C...
C...  INITIALIZE THE RUN COUNTER
      NORUN=0
C...
C...  BEGIN A RUN
1     NORUN=NORUN+1
C...
C...  INITIALIZE THE RUN TERMINATION VARIABLE
      NSTOP=0
C...
C...  READ THE FIRST LINE OF DATA
C...
      READ(NI,1000,END=999) (TITLE(I), I = 1, 20)
C...
C...  TEST FOR  END OF RUNS  IN THE DATA
C...
      DO 2 I = 1, 3
      IF(TITLE(I) .NE. ENDRUN(I)) GO TO 3
2     CONTINUE
C...
C...  AN END OF RUNS HAS BEEN READ, SO TERMINATE EXECUTION
999   STOP
```

Program MASKMAIN.FOR

```
C...
C...  READ THE SECOND LINE OF DATA
C...
3     READ(NI,*,END=999) TO, TF, TP
C...
C...  READ THE THIRD LINE OF DATA
C...
      READ(NI,*,END=999) NEQN, ERROR
C...
C...  PRINT A DATA SUMMARY
      WRITE(NO,1003)NORUN,(TITLE(I), I = 1, 20),
     1                    TO, TF, TP,
     2                    NEQN, ERROR
      WRITE(*,1003) NORUN, (TITLE(I), I = 1, 20),
     1                    TO, TF, TP,
     2                    NEQN, ERROR

C...
C...  INITIALIZE TIME
      T = TO
C...
C...  SET THE INITIAL CONDITIONS
      CALL INITIAL
C...
C...  SET THE INITIAL DERIVATIVES (FOR POSSIBLE PRINTING)
      CALL DERV
C...
C...  PRINT THE INITIAL CONDITIONS
      CALL PRINT(NI, NO)
C...
C...  SET THE INITIAL CONDITIONS FOR SUBROUTINE RKF45
      TV = TO
      DO 5 I = 1, NEQN
      YV(I) = Y(I)
5     CONTINUE
C...
C...  SET THE PARAMETERS FOR SUBROUTINE RKF45
C...
      RELERR = ERROR
      ABSERR = ERROR
      IFLAG = 1
      TOUT = TO + TP
C...
C...  CALL SUBROUTINE RKF45 TO START THE SOLUTION FROM THE INITIAL
C...  CONDITION (IFLAG = 1) OR COMPUTE THE SOLUTION TO THE NEXT PRINT
C...  POINT (IFLAG = 2)
4     CALL RKF45(FCN,NEQN,YV,TV,TOUT,RELERR,ABSERR,IFLAG,WORK,IWORK)
C...
C...  PRINT THE SOLUTION AT THE NEXT PRINT POINT
      T=TV
      TOUT = TV + TP
      PRINT *,"Time = ", T
      DO 6 I = 1, NEQN
      Y(I) = YV(I)
6     CONTINUE
      CALL DERV
      CALL PRINT(NI,NO)
      IF(IFLAG .EQ. 4 .OR. IFLAG .EQ. 7) IFLAG = 2
C...
C...  TEST FOR AN ERROR CONDITION
      IF(IFLAG .NE. 2) THEN
C...
C...     PRINT A MESSAGE INDICATING AN ERROR CONDITION
         WRITE(NO,1004) IFLAG
C...
```

Program MASKMAIN.FOR

```
C...      GO ON TO THE NEXT RUN
          GO TO 1
      END IF
C...
C...  CHECK FOR A RUN TERMINATION
      IF(NSTOP .NE. 0) GO TO 1
C...
C...  CHECK FOR THE END OF THE RUN
C...
      IF(TV .LT. (TF - 0.5D0*TP)) GO TO 4
C...
C...  THE CURRENT RUN IS COMPLETE, SO GO ON TO THE NEXT RUN
      GO TO 1
C...
C...  ******************************************************************
C...
C...  FORMATS
C...
1000  FORMAT(20A4)
1001  FORMAT(3E10.0)
1002  FORMAT(I5,20X,E10.0)
1003  FORMAT(1H1,
     1 ' RUN NO. - ',I3,2X,20A4,//,
     2 ' INITIAL T - ',E10.3,//,
     3 '   FINAL T - ',E10.3,//,
     4 '   PRINT T - ',E10.3,//,
     5 ' NUMBER OF DIFFERENTIAL EQUATIONS - ',I5,//,
     6 ' MAXIMUM INTEGRATION ERROR - ',E10.3,//,
     7 1H1)
1004  FORMAT(1H ,//,' IFLAG = ',I3,//,
     1 ' INDICATING AN INTEGRATION ERROR, SO THE CURRENT RUN'     ,/,
     2 ' IS TERMINATED.  PLEASE REFER TO THE DOCUMENTATION FOR'   ,/,
     3 ' SUBROUTINE',//,25X,'RKF45',//,
     4 ' FOR AN EXPLANATION OF THESE ERROR INDICATORS'            )
      END
      SUBROUTINE FCN(TV,YV,YDOT)
C...
C...  SUBROUTINE FCN IS AN INTERFACE ROUTINE BETWEEN SUBROUTINES RKF45
C...  AND DERV
C...
C...  NOTE THAT THE SIZE OF ARRAYS Y AND F IN THE FOLLOWING COMMON AREA
C...  IS ACTUALLY SET BY THE CORRESPONDING COMMON STATEMENT IN MAIN
C...  PROGRAM HEADHIT
      IMPLICIT DOUBLE PRECISION (A-H), DOUBLE PRECISION (O-Z)
      INTEGER NEQN, NSTOP, NORUN
      COMMON/T/          T,  XT,   NSTOP,     NORUN
     1      /Y/      Y(500)
     2      /F/      F(500)
C...
C...  THE NUMBER OF DIFFERENTIAL EQUATIONS IS AVAILABLE THROUGH COMMON
C...  /N/
C...
      COMMON/N/      NEQN
C...
C...  ABSOLUTE DIMENSION THE DEPENDENT VARIABLE, DERIVATIVE VECTORS
      DOUBLE PRECISION YV(500), YDOT(500)
C...
C...  TRANSFER THE INDEPENDENT VARIABLE, DEPENDENT VARIABLE VECTOR
C...  FOR USE IN SUBROUTINE DERV
C...
      T = TV
      DO 1 I = 1, NEQN
      Y(I) = YV(I)
1     CONTINUE
C...
```

Program MASKMAIN.FOR

```
C...  EVALUATE THE DERIVATIVE VECTOR
C...
      CALL DERV
C...
C...  TRANSFER THE DERIVATIVE VECTOR FOR USE BY SUBROUTINE RKF45
C...
      DO 2 I = 1, NEQN
      YDOT(I) = F(I)
2     CONTINUE
      RETURN
      END
```

Program MASKSUBS.FOR

```
C...  DECK MASKSUBS.FOR - SUBROUTINES REQUIRED TO IMPLEMENT A MODEL OF AN
C...  AVIATOR'S ORONASAL MASK.
C...
      SUBROUTINE INITIAL
C...
C...   Subroutine Initial sets initial conditions and defines constants
C...   passed to other modules in COMMON.  It is only called once.
C...
C...  DETAILED EXPLANATION OF THE EQUATIONS.
C...
C...
C...  NOTE:  The subscript notation indicated partial derivative WRT the
C...         subscript  E.G.  Xt ==> The first partial derivative of X
C...         WRT time.
C...
C...  This model estimates the flow and pressure within the mask hose and
C...  the oronasal cavity of the aviator's breathing mask during the
C...  breathing cycle.  The forcing function for the model is the volume
C...  change generated by the lung during breathing and pressure changes.
C...
C...     The present coding models the lung volume change by the following
C...  sinusoidal function.
C...
C...      V(t) = VLO - Qmax/w*(COS(wt) - 1), so that the lung flow is
C...
C...      Q(t) = Qmax*SIN(wt).
C...
C...      The airway resistances are modeled by simple quadratic fits to
C...  physiologic data.  The resistance of the bronchi and trachea are
C...  included in the lung model, but the resistance of the oronasal cavity
C...  is included in the mask model for convenience.  The oronasal resis-
C...  tances are accounted for separately for the nose and mouth by the
C...  following model.
C...
C...      deltaP =  K1*Q + K2*Q^2
C...
C...      The last term in the above accounts for inertance.
C...
C...      For the mouth,
C...
C...      K1 = 2.4 cm-H2O*sec/liter and K2 = 0.3 cm-H2O*sec^2/liter^2.
C...
C...      For the nose there are seperate K2's for inspiration and
C...  expiration.
C...
C...      K1 = 3.0 cm-H2O*sec/liter and
C...
C...      Inspiratory K2 = 3.0 cm-H2O*sec^2/liter^2.
C...
C...      Expiratory  K2 = 4.0 cm-H2O*sec^2/liter^2.
C...
C...      The overall pressure drop produced by the oronasal cavity
C...  is modelled by assigning the relative fraction of the total breathing
C...  flow to the mouth and nose respectively.  The individual drops are
C...  weighted by the flow fractions.  Thus,
C...
C...       Fm + Fn = 1.0, and deltaP = Fm*deltaPm + Fn*deltaPn.
C...
C...       The mask is modelled as a dead space and two variable area
C...  orifices through which inspiratory and expiratory flows separately
C...  pass.  The inspiratory valve model estimates the flow between
C...  the mask supply hose and the oronasal cavity during inspiration.
C...  The expiratory valve model relates the flow and pressure drop between the
C...  mask cavity and the external ambient atmosphere during expiratory flow.
C...  Both valve models are based on empirical data collected on the RAF
```

```
C...  P/Q mask, which meets the ASCC flow resistance standard.  Other valve
C...  models can be substituted.  The general form of the model is
C...
C...      Q = F(A,deltaP),
C...
C...  where A is the mask valve area and Q is the instantaneous
C...  flow through the valve.  The valve area, A, is in turn a function
C...  of the pressure drop, deltaP, across the valve.  It is solved by
C...  estimating the area of the valve from the pressure drop across the
C...  valve and then computing the flow through the valve using ideal
C...  orifice equations.  Mask leaks can be modelled by parallel flow
C...  paths to ambient, but none are included in this version.  Expiratory
C...  valve compensation is simulated by adding any positive difference
C...  between mask hose pressure and ambient pressure to the expiratory
C...  valve down stream pressure which is normally ambient pressure.
C...
C...  Mask hose and connector losses are modelled as simple tubes with
C...  flow-pressure drop relationships based on empirical data.  The
C...  regulator outlet flow-pressure relationship is based on curve fits
C...  to empirical data.
C...
C...
C...
C...  ODE COMMON
C...
C...   /Y/ time variables
C...   /F/ time derivatives of variables
C...   /S/ spatial derivatives of variables
C...   /R/ & /I/ real and integer parameters required to define constants and
C...           define the spatial integration grid.
C...
      PARAMETER (NDE=3)
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      DOUBLE PRECISION  KG, KP1, KP1OK, KOKM1, KOKP1
      INTEGER NSTOP, NORUN, IP, NEQN
      COMMON/T/         T, XT,    NSTOP,     NORUN
     1      /Y/ VOL(NDE), FLOW(NDE), PRESS(NDE),         !
C...
C...            VOL(1)  [=]  Cumulative volume flow from regulator  [m^3]
C...            VOL(2)  [=]  Cumulative volume flow into mask       [m^3]
C...            VOL(3)  [=]  Cumulative Respired volume (Insp - Exp)[m^3]
C...
C...            FLOW(1) [=]  Instantaneous flow from regulator [m^3/sec]
C...            FLOW(2) [=]  Instantaneous mask flow        [m^3/sec]
C...            FLOW(3) [=]  Instantaneous Oro-nasal flow  [m^3/sec]
C...
C...            PRESS(1) [=]  Regulator Outlet Pressure  [Pa]
C...            PRESS(2) [=]  Mask/Oronasal Cavity Press.[Pa]
C...            PRESS(3) [=]  Intra Pulmonary Pressure   [Pa]
C...
     2      /F/ DVDT(NDE), DFDT(NDE), DPDT(NDE),
C...
C...        Derivatives of Volumes and Pressures
C...
     3      /S/ DUMMY                    !   Spatial Derivatives if needed
     4      /R/ PI, PO, VO, TO, RUNIV,   ! Pi = 3.1412..., Std P,V,T& R
     *          PA2MM,
     *          KG, KP1, KOKM1, KP1OK, KOKP1,! K, K/(K-1), (K+1)/K, 1/K, 2/K, etc
     *          TWOOK, TMKOK, TWOOKP1,   ! Adiabatic constant and
     *          TWOOKM1, ONEOK,          ! derived constants
     *          CD,                      ! Orifice Discharge Coefficient
     *          TBODY, TAMB, PAMB,       ! Temperatures - Body & Ambient
     *          GMW, GMWS, GASK, GASKS, ! Gas parameters MW & Specif Heat ratios
     *          GAM, OMEGA, TINSP, FRM, ! Breathing flow parameters
     *          TR, TPAUSE, RR, VTIDAL, ! Breathing flow parameters
```

Program MASKSUBS.FOR

```
     *          VFRCO, FRC, PPWBODY,    ! Initial, Current FRC, ppH2O Body
     *          AM(2), AMI(3), AME(3),  ! Airway parameters
     *          VAO, VMAO, CINERT, DPAW,! Airway parameters
     *          AWI, DVL,               ! Airway inertance
     *          PREGO, PREF, Gdot1,     ! Regulator Outlet Pressure
     *          VDSMSK, AMIV, AMEV,     ! Mask parameters
     *          QIV, QEV,               ! Mask valve flow indicators
     *          VOL1, VOL2, VOL3, VOL4, ! Volumes
     *          AWH, AWO, AWN, AWAr,    ! Atomic wts
     *          GMN2, GMO2, GMCO2,      ! Molecular wts
     *          GMH2O, GMWAIR,          !    "      "
     *          FINSP(4), FEXP(4)       !Gas Fractions, O2,N2+Ar,CO2,H2O
     5      /I/ IP , IRELF              ! Print Counter
     6      /N/ NEQN
C...
C...  Set Constants employed in simulation
C...
C...  Pi = 3.14159...
C...  TO = Freezing point of water at 1 atmosphere
C...  VO = Volume occupied by 1 kg-mol of ideal gas at 1 atm and Tzero
C...  PO = Standard atmospheric pressure in Pascals [N/m^2]
C...
      Pi = DACOS(-1.D0)
      TO = 273.1D0           ! deg K
      VO = 22.4097D0         ! m^3/kg-mol
      PO = 1.01325D5         ! Pa
      GASK = 1.4D0           ! k = Specific Heat Ratio Cp/Cv
      CD = 0.68              ! Orifice coefficient
      FM  = 0.75D0           ! Fraction mouth breathing
      VAO = 150.D-6          ! Respiratory Anatomical Dead Space (m^3)
      VDSMSK = 150.D-6       ! Oronasal Mask Dead Space (m^3)
      VFRCO = 2.5D-3         ! Resting Functional Residual Capacity [m^3]
      DVL = 2.2502D-7        ! Rate of Lung Volume Increase per Pascal of PPB
      GMWH = 1.008D0         ! Atomic Wt Hydrogen
      AWN = 14.01D0          ! AW Nitrogen
      AWO = 16.D0            ! AW Oxygen
      AWC = 12.01D0          ! AW Carbon
      AWAr = 39.94D0         ! AW Argon
C...
C...  DERIVED CONSTANTS
C...
C...  PV = RT  Universal Gas Law
C...
C...  RUNIV = PO*VO/TO  Universal Gas Constant
C...
      RUNIV = PO*VO/TO
      KP1 = GASK + 1.D0              ! k + 1
      KOKM1 = GASK/(GASK - 1.D0)     ! k/(k - 1)
      KP1OK = KP1/GASK               ! (k + 1)/k
      KOKP1 = GASK/KP1               ! k/(k + 1)
      TWOOK = 2.D0/GASK              ! 2/k
      TMKOK = (2.D0 - GASK)/GASK     ! (2 - k)/k
      TWOOKP1 = 2.D0/KP1             ! 2/(k + 1)
      TWOOKM1 = 2.D0/(GASK - 1.D0)   ! 2/(k - 1)
      ONEOK   = 1.D0/GASK            ! 1/k
      TAMB    = TO + 25.D0           ! Ambient Temperature
      TBODY =  TO + 37.D0            ! Body Temperature
      PA2MM = 760.D0/1.01325D5       ! Conversion Factor Pa to mmHg
      PAMB  = 1.01325D5
      PREF  = PAMB
      GMH2O = AWH*2.D0 + AWO         ! MW Water
      GMCO2 = AWC + 2.D0*AWO         ! MW Carbon Dioxide
      GMO2  = 2.D0*AWO               ! MW N2
      GMN2  = 2.D0*AWN               ! MW O2
      GMWAIR = 0.2095D0*GMO2
```

```
     *        + 0.7808D0*GMN2
     *        + 0.0093D0*AMAr          ! MW Dry Air (Neglect 0.04% CO2)
      AWR = ((VAO + VDSMSK)*3.D0/4.D0/Pi)**(1.D0/3.D0) ! Airway radius
      AWL = 2.D0*AWR        ! Characteristic length of airway
      AWA = Pi*AWR*AWR ! Characteristic airway area
      AWI = AWL/AWA         ! Airway Inertance (m^-1)
C...
C...
C...  INITIAL CONDITIONS (T = 0)
C...
C...   RESPIRATION RATE (Breaths/min)
C...
      RR = 20.D0/60.D0      !  Breaths per second
      TR = 1.D0/RR          !  Respiratory period
      TINSP = TR/2.D0       !  Inspiratory period
      TPAUSE = 0.1D0*TR     !  Interbreath pause duration
C...
C...   VDOTE = MINUTE VOLUME (M^3/SEC]
C...
      VTIDAL = 1.5D-3       !  Tidal Volume = 1.5 liter/breath
      VDOTE = RR*VTIDAL     !  VdotE in m^3/sec
      QAM = VDOTE*Pi        !  Peak Flow from sinusoidal flow profile
      OMEGA = 2.D0*Pi*RR    !  RR in radians/sec
      FRM = 0.75D0          !  Fraction Mouth Breathing
      FINSP(1) = 0.5D0      !  Fr O2 inspired
      FINSP(2) = 0.5D0      !  Fr N2 (Inert) inspired
      FINSP(3) = 0.0D0      !  Fr CO2 inspired
      FINSP(4) = 0.0D0      !  Fr H2O inspired
      FEXP(1) = 0.17D0      !  Fr O2 expired
      FEXP(3) = 0.04D0      !  Fr CO2 expired
      FEXP(4) = 0.062D0     !  Fr H2O expired
      FEXP(2) = 1.D0 - (FEXP(1) + FEXP(3) + FEXP(4))  ! Fr N2 (Inert) expired
      VOL(1) = Pi*((2.54D-2)/2.D0)**2.D0*2.D0 ! 2 m of 1 in ID hose
      VOL1 = VOL(1)           !  Parameterize initial volumes
      VOL(2) = VDSMSK + VAO !  Include mask cavity and anatomical dead space
      VOL2 = VOL(2)
      VOL(3) = VFRCO
      VOL3 = VOL(3)
      PREGO = PAMB            ! Initialize regulator outlet pressure
      AMIV = 0.D0
      AMEV = 0.D0
      QIV = 0.D0
      QEV = 0.D0
      PPWBODY = 47.D0/PA2MM ! Convert saturation pressure to Pascals
      i = 1
      DO WHILE (i .LE. NDE)! Initialize flows, pressures, and derivatives
         PRESS(i) = PAMB
         FLOW(i)  = 0.D0
         DFDT(i)  = 0.D0
         DVDT(i)  = 0.D0
         DPDT(i)  = 0.D0
         i = i + 1
      END DO
C...
C...  Compute Starting values for derivatives by calling DERV
C...
      CALL DERV
      IP = 0        !  Initialize print flag
      IRELF = 0  !  Initialize RD flag
      RETURN
      END
C...
      SUBROUTINE DERV
C...
C...  DERV CALCULATES THE TIME DERIVATIVES TO BE INTEGRATED BY RKF45
```

Program MASKSUBS.FOR

```
C...
C...  ODE COMMON
C...
C...   /Y/ time variables
C...   /F/ time derivatives of variables
C...   /S/ spatial derivatives of variables
C...   /R/ & /I/ real and integer parameters required to define constants and
C...            define the spatial integration grid.
C...
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      DOUBLE PRECISION  KG, KP1, KP1OK, KOKM1, KOKP1
      DOUBLE PRECISION  DPMIE, DPMI, DPME
      INTEGER NSTOP, NORUN, NDE, IP, NEQN
      PARAMETER (NDE=3)
      COMMON/T/          T, XT,    NSTOP,     NORUN
     1      /Y/ VOL(NDE), FLOW(NDE), PRESS(NDE),        !
C...
C...           VOL(1)  [=]  Cumulative volume flow from regulator
C...           VOL(2)  [=]  Cumulative volume flow into mask
C...           VOL(3)  [=]  Cumulative Respired volume (Insp - Exp)
C...
C...           FLOW(1) [=]  Instantaneous flow from regulator [m^3/sec]
C...           FLOW(2) [=]  Instantaneous mask flow        [m^3/sec]
C...           FLOW(3) [=]  Instantaneous Oro-nasal flow  [m^3/sec]
C...
C...           PRESS(1) [=]  Regulator Outlet Pressure
C...           PRESS(2) [=]  Mask hose pressure
C...           PRESS(3) [=]  Mask cavity Pressure = Intraoral pressure
C...
     2      /F/ DVDT(NDE), DFDT(NDE), DPDT(NDE),
C...
C...      Derivatives of Volumes and Pressures
C...
     3      /S/ DUMMY                    !   Spatial Derivatives if needed
     4      /R/ PI, PO, VO, TO, RUNIV,   ! Pi = 3.1412..., Std P,V,T& R
     *         PA2MM,
     *         KG, KP1, KOKM1, KP1OK, KOKP1,! K, K/(K-1), (K+1)/K, 1/K, 2/K, etc
     *         TWOOK, TMKOK, TWOOKP1,   ! Adiabatic constant and
     *         TWOOKM1, ONEOK,          ! derived constants
     *         CD,                      ! Orifice Discharge Coefficient
     *         TBODY, TAMB, PAMB,       ! Temperatures - Body & Ambient
     *         GMW, GMWS, GASK, GASKS,  ! Gas parameters MW & Specif Heat ratios
     *         QAM, OMEGA, TINSP, FRM,   ! Breathing flow parameters
     *         TR, TPAUSE, RR, VTIDAL,  ! Breathing flow parameters
     *         VFRCO, FRC, PPWBODY,     ! Initial, Current FRC, ppH2O Body
     *         AM(2), ANI(3), ANE(3),   ! Airway parameters
     *         VAO, VMAO, CINERT, DPAW,! Airway parameters
     *         AWI, DVL,                ! Airway parameters
     *         PREGO, PREF, Gdot1,      ! Regulator Outlet Pressure
     *         VDSMSK, AMIV, AMEV,      ! Mask parameters
     *         QIV, QEV,                ! Mask valve flow indicators
     *         VOL1, VOL2, VOL3, VOL4,  ! Volumes
     *         AWN, AWO, AWH, AWAr,     ! Atomic wts
     *         GMN2, GMO2, GMCO2,       ! Molecular wts
     *         GMH2O, GMWAIR,           !     "      "
     *         FINSP(4), FEXP(4)        !Gas Fractions, O2,N2+Ar,CO2,H2O
     5      /I/ IP , IRELF            ! Print Counter
     6      /N/ NEQN
C...
C...  THE NUMBER OF DIFFERENTIAL EQUATIONS IS IN COMMON/N/ FOR USE IN
C...  SUBROUTINE FCN
C...
C...  COMPUTE DERIVATIVES
C...
C...   Define some statement Functions
```

Program MASKSUBS.FOR

```
C...
C...  Inspiratory and expiratory resistance of the mouth
C...
      DPMIE(Q) = 2.35D5*Q + 2.94D7*Q*Q ! Mouth dP vs Q for inspiration and expiration
C...                                  ! Volume Flow
C...  Inspiratory nose resistance
C...
      DPNI(Q)  = 2.94D5*Q + 2.94D8*Q*Q  ! Nose  dP vs Q for inspiration
C...                                    ! Volume flow
C...  Expiratory nose resistance
C...
      DPNE(Q)  = 2.94D5*Q + 3.92D8*Q*Q ! Nose  dP vs Q for expiration
C...                                   ! volume flow
      VTOT = VOL2 + VOL(3)  ! Lung Volume and Deadspace Volume
      TRD = TR + TPAUSE + TPAUSE/2.D0
      TC = TR + TPAUSE      ! Time per respiratory cycle
      RDF = 0.D0  ! RD Flag OFF
      IF(T .GE. TRD .AND. PAMB .GT. 6.68608D4) THEN   !RAPID DECOMPRESSION
          PAMB = DMAX1(1.01325D5
     *    - 2.D0*3.44643D4*(T - TRD)/TC, 6.68607D4)
          RDF = 1.D0  ! Set RD Flag ON
      END IF
      PREF = PAMB
      IF (PRESS(1) - PREF .GT. 5.D2 .AND. FLOW(1) .EQ. 0.D0) THEN
          PRESS(1)   = DMIN1(PREF + 5.D2, PRESS(1))! Simulate a 2 InWg relief valve
          PREGO = PRESS(1)
        ELSE
          PREGO = DMIN1(PREGO, PREF)
          PRESS(1) = DMIN1(PREGO,PRESS(1))
      END IF
      RHOAO = PRESS(2)*GMWAO/RUNIV/TBODY    ! Density Oronasal cavity
      IF (DMOD(T,TC) - TPAUSE .GE. 0.D0) THEN
           XT = DMOD(T,TC) - TPAUSE     ! Compute time from onset of
         ELSE                           ! Inspiration
           XT = 0.D0                    ! Pause time
      END IF
      IF(T .GT. TRD .AND. T .LT. 2.D0*TC + TPAUSE/2.D0) XT = 0.D0
      QI = 0.D0   ! Inspiratory Flag
      IF(XT .GT. 0.D0 .AND. XT .LE. TINSP) QI = 1.D0  ! Inspiratory Flag
      QE = 1.D0 - QI                                  ! Expiratory Flag
      i = 1
      DO WHILE (i .LE. NDE)
       PRESS(i) = DMAX1(PRESS(i), 1.D2)! Absolute pressure .GE. 100 Pa
         i = i + 1      ! Don't allow negative absolute pressure
      END DO
      IF(PRESS(2) .GT. 0.D0) FINSP(4) = PPWBODY/PRESS(2)   ! Mole Fraction H2O
      FINSP(2) = 1.D0 - (FINSP(1) + FINSP(3) + FINSP(4))   ! Inert Gas Fraction
      GMWAO = CALMW(FINSP)                    ! MW Oronasal Cavity Inspiration
      RHOAO = PRESS(3)*GMWAO/RUNIV/TBODY      ! Density Oronasal cavity
      GMWMH = FINSP(1)*GMO2 + FINSP(2)*GMN2 ! MW Mask Hose
      RHOMH = PRESS(1)*GMWMH/RUNIV/TAMB       ! Density Mask Hose [kg/m^3]
      OMF = 1.D0 - FRM ! Fraction Nose Breathing = 1 - Fraction Mouth Breathing
      IF( XT .GT. 0.D0 ) THEN
          DFDT(3) = QAM*OMEGA*DCOS(OMEGA*XT) ! Lung Volume Flow Derivative
          FLOW(3) = QAM*DSIN(OMEGA*XT)       ! Lung Volume Flow  [m^3/sec]
          Gdot3 = FLOW(3)*RHOAO              ! Lung Mass Flow [kg/sec]
          DP = DABS(FRM*DPMIE(FLOW(3)) + OMF*DPNI(FLOW(3))*QI ! Pressure
     *              + OMF*DPNE(FLOW(3))*QE)    ! loss through nose & mouth
          PRESS(3) = PRESS(2) - DP*QI + DP*QE    ! lung pressure
          C3 = VTOT*GMWAO/RUNIV/TBODY  ! Lung Capacitance
          IF(Gdot3 .NE. 0.D0) THEN
             R3 = DP/Gdot3    ! Airway Resistance
            ELSE
             R3 = 0.D0
          END IF
```

```
          DPDT(3) =  -(1.D0/C3*Gdot3 + DFDT(3)*R3) !  dP/dt for Lung
        ELSE
          FLOW(3) = 0.D0
          Gdot3 = 0.D0
          DVDT(3) = 0.D0
          DPDT(3) = 0.D0
          IF(FLOW(2) .EQ. 0.D0) PRESS(3) = PRESS(2)
          RHOAO = PRESS(2)*GMWAO/RUNIV/TBODY    ! Density Oronasal cavity
      END IF
      DPLUNG = PRESS(3) - PAMB  ! Differential Lung Pressure
C
C  Calculate the area of the inspiratory and expiratory valves
C  from curve fits to RAF P/Q Mask valve data (ASCC compliant)
C
C          !  Exp valve has 0.5 InWg cracking Pressure
       IF(PRESS(2) .GT. (PRESS(1) + 50.D0) ) THEN
          QIV = 0.D0
          PRESS(1) = DMAX1(PREGO, PRESS(1))
          RHOMH = PRESS(1)*GMWMH/RUNIV/TAMB  ! Density Mask Hose [kg/m^3]
          FLOW(1) = 0.D0
          Gdot1 = 0.D0
          DPDT(1) = 0.D0
          FEXP(2) = 1.D0 - (FEXP(1) + FEXP(3) + FEXP(4))  ! Inert gas fraction
          GMWE = CALMW(FEXP)   ! Expired gas molecular weight
          PEXPV = PAMB + DMAX1(0.D0, PRESS(1) - PAMB) + 50.D0 ! 0.5 Inch Wg Spring Pressure
               ! Compensation pressure
               ! Expiratory valve back pressure can be tailored to
               ! compensation characteristics of a particular valve
          DPX  = DMAX1(PRESS(2) - PEXPV,0.D0)  ! Delta-P Exp Valve
          AMEV = DMAX1(0.D0, DMIN1(3.84D-7*DPX -1.11D-5,1.6D-4))!Area Expiratory Valve
C                            PMASK     PMHOSE
          CALL ORIFLO(CD, AMEV, PRESS(2), PEXPV , TBODY, GMWE,
     *                FLOW(2), Gdot2, PCRIT)    ! Expiratory valve flow
          FLOW(2) = DMIN1(-FLOW(2), 0.D0)  ! Flow is reversed for expiration
          IF(FLOW(2) .LT. 0.D0) THEN
             QEV = 1.0
            ELSE
             QEV = 0.D0
          END IF
          DPDT(2) = PRESS(2)/VTOT*(FLOW(2) - FLOW(3))  ! Mask dP/dt.
        ELSEIF(PRESS(1) .GT. (PRESS(2) + 10.D0)) THEN
C           IF (FLOW(1) .NE. 0.D0) THEN
C              PRESS(1) = DMIN1(PRESS(1),PREF)
C             ELSE
C              PRESS(1) = DMIN1(PRESS(1), PREGO)
C           END IF
          DPI  = PRESS(1) - PRESS(2) !Delta-P Inspiratory Valve
          AMIV = DMAX1(0.D0, DMIN1(1.009D-6*DPI,1.5D-4)) !Area Insp Valve
          GMWI = CALMW(FINSP) ! Inspiratory gas molecular weight
C                             PMHOSE    PMASK
          CALL ORIFLO(CD, AMIV, PRESS(1), PRESS(2), TAMB, GMWI,
     *                FLOW(2), Gdot2, PCRIT)   ! Inspiratory valve flow
          FLOW(2) = DMAX1(FLOW(2), 0.D0)
          IF(FLOW(2) .GT. 0.D0) THEN
             QIV = 1.0
            ELSE
             QIV = 0.D0
          END IF
          RHOAO = PRESS(2)*GMWAO/RUNIV/TBODY    ! Density Oronasal cavity
          RHOMH = PRESS(1)*GMWMH/RUNIV/TAMB     ! Density Mask Hose [kg/m^3]
          Gdot1 =  FLOW(2)*RHOAO ! Mass flow into mask
          FLOW(1) = Gdot1/RHOMH  ! Volume flow from mask tube
          DPH = DPHOSE(Gdot1)    ! Pressure loss in hose
          PREGO = DMIN1(PRESS(1) + DPH, PREF) ! Regulator outlet pressure
          DELTAP = PREGO - PREF  ! Differential pressure relative to reference
```

Program MASKSUBS.FOR

```
          DPDT(1) = 1.D0/VOL1*(PREGO*GTREG(DELTAP)/RHOMH    ! GTREG is mass flow v DP
     *             - PRESS(1)*FLOW(1)) ! dP/dt for mask hose outlet pressure
          DPDT(2) = PRESS(2)/VTOT*(FLOW(2) - FLOW(3)) ! dP/dt for Mask
        ELSE
          IF( QIV .EQ. 0.D0 ) THEN
             FLOW(1) = 0.D0
             Gdot1   = 0.D0
             DPDT(1) = 0.D0
             DFDT(1) = 0.D0
             PREGO = PREF
             PRESS(1) = DMIN1(PREGO, PRESS(1))
             RHOMH = PRESS(1)*GMWMH/RUNIV/TAMB     ! Density Mask Hose [kg/m^3]
          END IF
          IF(QIV + QEV .EQ. 0.D0) THEN
            FLOW(2) = 0.D0
            DFDT(2) = 0.D0
            IF(FLOW(3) .EQ. 0.D0) PRESS(3) = PRESS(2)
            DPDT(2) = -PRESS(2)/VTOT*FLOW(3) ! dP/dt for Mask
          END IF
      END IF
      DVDT(1) = FLOW(1)  ! Cumulative volume flow from regulator
      DVDT(2) = FLOW(2)  ! Flow is derivative of volume
      DVDT(3) = FLOW(3)  ! Flow is derivative of volume
 1000 RETURN
      END
      SUBROUTINE PRINT(NI,NO)
C...
C...  Prints selected output variables at the specified time interval.
C...
C...
C...  ODE COMMON
C...
C...   /Y/ time variables
C...   /F/ time derivatives of variables
C...   /S/ spatial derivatives of variables
C...   /R/ & /I/ real and integer parameters required to define constants and
C...            define the spatial integration grid.
C...
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      PARAMETER (NDE=3)
      DOUBLE PRECISION  KG, KP1, KP1OK, KOKM1, KOKP1
      INTEGER NSTOP, NORUN,  IP, NEQN
      COMMON/T/         T, XT,    NSTOP,     NORUN
     1      /Y/ VOL(NDE), FLOW(NDE), PRESS(NDE),       !
C...
C...           VOL(1)  [=]  Cumulative volume flow from regulator
C...           VOL(2)  [=]  Cumulative volume flow into mask
C...           VOL(3)  [=]  Cumulative Respired volume (Insp - Exp)
C...
C...           FLOW(1) [=]  Instantaneous flow from regulator [m^3/sec]
C...           FLOW(2) [=]  Instantaneous mask flow       [m^3/sec]
C...           FLOW(3) [=]  Instantaneous Oro-nasal flow  [m^3/sec]
C...
C...           PRESS(1) [=]  Regulator Outlet Pressure
C...           PRESS(2) [=]  Mask hose pressure
C...           PRESS(3) [=]  Mask cavity Pressure = Intraoral pressure
C...
     2      /F/ DVDT(NDE), DFDT(NDE), DPDT(NDE),
C...
C...       Derivatives of Volumes and Pressures
C...
     3      /S/ DUMMY                     !   Spatial Derivatives if needed
     4      /R/ PI, PO, VO, TO, RUNIV,  ! Pi = 3.1412..., Std P,V,T& R
     *         PAZMM,
     *         KG, KP1, KOKM1, KP1OK, KOKP1,! K, K/(K-1), (K+1)/K, 1/K, 2/K, etc
```

```
     *           TWOOK, THKOK, TWOOKP1,  ! Adiabatic constant and
     *           TWOOKM1, ONEOK,         ! derived constants
     *           CD,                     ! Orifice Discharge Coefficient
     *           TBODY, TAMB, PAMB,      ! Temperatures - Body & Ambient
     *           GMW, GMWS, GASK, GASKS, ! Gas parameters MW & Specif Heat ratios
     *           QAM, OMEGA, TINSP, FRM, ! Breathing flow parameters
     *           TR, TPAUSE, RR, VTIDAL, ! Breathing flow parameters
     *           VFRCO, FRC, PPWBODY,   ! Initial, Current FRC, ppH2O Body
     *           AM(2), ANI(3), ANE(3),  ! Airway parameters
     *           VAO, VMAO, CINERT, DPAW,! Airway parameters
     *           AWI, DVL,               ! Airway parameters
     *           PREGO, PREF, Gdot1,     ! Regulator Outlet Pressure
     *           VDSMSK, AMIV, AMEV,     ! Mask parameters
     *           QIV, QEV,               ! Mask valve flow indicators
     *           VOL1, VOL2, VOL3, VOL4, ! Volumes
     *           AWN, AWO, AWH, AWAr,    ! Atomic wts
     *           GMN2, GMO2, GMCO2,      ! Molecular wts
     *           GMH2O, GMWAIR,          !    "     "
     *           FINSP(4), FEXP(4)       !Gas Fractions, O2,N2+Ar,CO2,H2O
     5      /I/ IP , IRELF             ! Print Counter
     6      /N/ NEON
C...
C...
C...  PRINT A HEADING FOR THE NUMERICAL SOLUTION
C...
        IP = IP + 1
C       WRITE(NO,2)
C       WRITE(NO,2) T
C...
C...  PRINT THE SOLUTION
C...
       WRITE( *,'(1X,2F10.3)') T, XT
       WRITE( *,3) T,  PREGO, (PRESS(k), k=1,NDE), (DPDT(k), k=1,NDE)
       WRITE( *,3) T,  (VOL(k), k=1,NDE),   (FLOW(k), k=1,NDE)
       WRITE(NO,2) T,  PAMB, PREGO, (PRESS(k), k=1,NDE),
     *   (DPDT(k), k=1,NDE), (VOL(k), k=1,NDE),   (FLOW(k), k=1,NDE)
 2    FORMAT(F8.4, 16E17.8)
 3    FORMAT(1X,F8.4, 30E13.5)
      RETURN
      END
      DOUBLE PRECISION FUNCTION GTREG(DP)
      DOUBLE PRECISION DP
C...
C...  REGULATOR MASS FLOW VS OUTLET PRESSURE
C...  BASED ON CURVE FIT
C...
      IF(DP .GE. 0.D0) THEN
         GTREG = 0.D0
         RETURN
      END IF
      GTREG = DMAX1(1.D-10,-1.78D-3 - 4.985D-5*DP - 1.585D-7*DP**2
     *              - 2.136D-10*DP**3)
      IF(DP .LT. -3.5D2) GTREG = 5.D-3
      RETURN
      END
      DOUBLE PRECISION FUNCTION POREG(Gt)
      DOUBLE PRECISION  Gt
C...
C...  Regulator differential outlet pressure = f(mass flow at outlet)
C...  Units:  Pa vs kg/sec
C...
      POREG = -46.72D0 - 1.219D4*Gt - 1.009D7*Gt*Gt  ! Pascals
      IF(Gt .LT. 0.D0) POREG = 0.D0
      RETURN
      END
```

Program MASKSUBS.FOR

```
      DOUBLE PRECISION FUNCTION DPHOSE(Gt)
C
C...  Mask hose Resistance
C...
      DOUBLE PRECISION Gt
      DPHOSE =      1.1462D4*Gt + 5.25D6*Gt*Gt ! delta-P hose vs volume flow
      RETURN
      END
      SUBROUTINE ORIFLO(CDORIF, AREA, PUP, PDWN, TABS, GMWT,
     *  FLOWVOL, FLOWMASS, PCRIT)
C
C**  Subroutine ORIFLO uses the compressible flow equations for an ideal
C     gas to compute the flow through an orifice given the upstream and
C     downstream conditions, the orifice area, and the gas parameters.
C
C
C    PARAMETERS           DESCRIPTION                          UNITS
C
C       CD      [=]  The orifice discharge coefficient         [UNITLESS]
C       AREA    [=]  The area of the orifice                   [SQ METERS]
C       TABS    [=]  The absolute temperature                  [DEG KELVIN]
C       GMW     [=]  The molecular weight of the gas           [KG/KG-MOL]
C       GASK    [=]  The adiabatic exponent ratio of specific
C                    heats of the gas (Cp/Cv)                  [UNITLESS]
C       RUNIV   [=]  Universal Gas Constant                    [NEWTON*METERS/KGMOL/DEG K]
C
c    VARIABLES
C
C       PUP     [=]  Upstream absolute pressure                [PASCALS]
C       PDwn    [=]  Downstream absolute pressure              [PASCALS]
C
C******OUTPUT PARAMETERS
C
C       FLOWVOL  [=] Volumetric flow                           [CU METERS/SEC]
C       FLOWMASS [=] Mass flow                                 [KG/SEC]
C       PCRIT    [=] Critical downstream pressure for sonic flow.[PASCALS]
C
C*********NOTE:  If other units are desired, the universal gas constant
C
C       RUNIV must be changed to the correct value for the chosen units.
C
      IMPLICIT DOUBLE PRECISION (A-H,O-Z)
      DOUBLE PRECISION  KG, KP1, KP1OK, KOKM1, KOKP1
      COMMON /R/ PI, PO, VO, TO, RUNIV, ! Pi = 3.1412..., Std P,V,T& R
     *         PA2MM,
     *         KG, KP1, KOKM1, KP1OK, KOKP1,! K, K/(K-1), (K+1)/K, 1/K, 2/K, etc
     *         TWOOK, TMKOK, TWOOKP1,  ! Adiabatic constant and
     *         TWOOKM1, ONEOK,         ! derived constants
     *         CD,                     ! Orifice Discharge Coefficient
     *         TBODY, TAMB, PAMB,            ! Temperatures - Body & Ambient
     *         GMW, GMWS, GASK, GASKS, ! Gas parameters MW & Specif Heat ratios
     *         QAM, OMEGA, TINSP, FRM,      ! Breathing flow parameters
     *         TR, TPAUSE, RR, VTIDAL, ! Breathing flow parameters
     *         VFRCO, FRC, PPWBODY,    ! Initial, Current FRC, ppH2O Body
     *         AM(2), ANI(3), ANE(3),  ! Airway parameters
     *         VAO, VMAO, CINERT, DPAW,! Airway parameters
     *         AWI, DVL,
     *         PREGO, PREF, Gdot1,     ! Regulator Outlet Pressure
     *         VDSMSK, AMIV, AMEV,     ! Mask parameters
     *         QIV, QEV,               ! Mask valve flow indicators
     *         VOL1, VOL2, VOL3, VOL4, ! Volumes
     *         AWN, AWO, AWH, AWAr,    ! Atomic wts
     *         GMN2, GMO2, GMCO2,      ! Molecular wts
     *         GMH2O, GMWAIR,          !     "      "
     *         FINSP(4), FEXP(4)       !Gas Fractions, O2,N2+Ar,CO2,H2O
```

```
      COMMON /I/ IP , IRELF
C
C  PRESERVE THE CALLING CONSTANTS AND VARIABLES.
C
C
C  THESE CONSTANTS DO NOT CHANGE DURING A SINGLE CALL,
C  BUT MAY CHANGE BETWEEN CALLS.
C
      IF(PUP .EQ. PDWN .OR. PUP .LE. 0.D0 .OR. TABS .LE. 0.D0 .OR.
     *   PDWN .LE. 0.D0 .OR. AREA .LE. 0.D0 .OR. GMWT .LE. 0.D0) THEN
        FLOWMASS = 0.0D0
        FLOWVOL  = 0.0D0
        RETURN
      END IF
      WT = GMWT
      PU = PUP
      PD = PDWN
      PR = PD/PU
      C  = CDORIF
      A  = AREA
      T  = TABS
      RGAS = RUNIV/WT               !  SPECIFIC GAS CONSTANT
      FWD = 1.D0                    !  FORWARD FLOW FLAG
      IF (PR .GT. 1.0D0) THEN
         FWD = -FWD                 !  REVERSE FLOW
         PR = 1.0D0/PR              !  EXCHANGE PU & PD
         PTEMP = PU
         PU = PD
         PD = PTEMP
      END IF
      RHOU = PU/(RGAS*T)            !  UPSTREAM GAS DENSITY
      PCRIT = TWOOKP1**KOKM1*PU     !  CRITICAL PRESSURE FOR SONIC FLOW
C
C** IF DOWNSTREAM CONDITIONS < PCRIT USE SONIC EQUATIONS
C
   99 FORMAT(5X,5E20.8)
      IF(PD .LT. PCRIT) THEN
C
C     ** SONIC CONDITIONS APPLY IF PD < .OR. = PCRIT
C
         FACT = 2.0D0*KOKP1
         ARG = PU*RHOU*FACT*TWOOKP1**TWOOKM1
         IF(ARG .LT. 0.D0) PAUSE 'INVALID ARGUMENT IN ORIFLOW'
         FLOWMASS = FWD*C*A*DSQRT(ARG)
         FLOWVOL  = FLOWMASS/RHOU
       ELSE
C
C     ** ELSE USE THE SUBSONIC EQUATIONS
C
         FACT = 2.0D0*KOKM1
         ST = DSQRT(T)
         ARG =  FACT/RGAS*(PR**TWOOK - PR**KP1OK)
         IF(ARG .LT. 0.D0) PAUSE 'INVALID ARGUMENT IN ORIFLOW'
         FLOWMASS = FWD*C*A/ST*PU*DSQRT(ARG)
         FLOWVOL = FLOWMASS/RHOU
      END IF
      RETURN
      END
      SUBROUTINE DICAN(IATOP,ALT,PABS,TEMPK)
C
C  MODIFIED 7/12/87 TO INCLUDE IMPROVEMENTS PROGRAMMED BY MR L. GILL OF
C  MOOG CARLETON GROUP.
C
C  DOUBLE PRECISION VERSION CREATED 12/1/93 FOR MASKMODEL
C
```

Program MASKSUBS.FOR

```
C  ICAN COMPUTES THE ABSOLUTE PRESSURE FOR A GIVEN ALTITUDE
C    (OR VICE VERSA) BY ICAN STANDARD ATMOSPHERE MODEL(1953).
C
C        TO COMPUTE PRESSURE FROM ALTITUDE SET IATOP .GE. 0
C        TO COMPUTE ALTITUDE FROM PRESSURE SET IATOP .LT. 0
C
C    UNITS ON ALTITUDE RETURNED IN FEET.
C
C    THE ABSOLUTE TEMPERATURE ESTIMATE FOR THE ALTITUDE IS
C    RETURNED IN DEG KELVIN(TEMPK).
C
C    THE MODEL IS MOST ACCURATE BETWEEN -1,000 AND 100,000 FEET.
C
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      INTEGER IATOP
      X = 5.256D0
      TC = 6.5D-3
      TO = 288.16D0
      SCALE = 1.0D-5
      IF(IATOP .LT. 0) GO TO 200
      ALT = ALT/1000.D0
C
C  CHECK FOR TROPOPAUSE
C
      IF(ALT .GT. 36.D0) GO TO 150
C
C  MODEL FOR BELOW TROPOPAUSE
C
      TEMPK = TO - TC*ALT
      PABS = ((TO-1.9812D0*ALT)/TO)**X+(2.15287D0*ALT-8.16554D0)*SCALE
      GO TO 999
C
C  MODEL FOR ABOVE TROPOPAUSE
C
  150 C = 0.224D0*10.D0**((36.0889D0 - ALT)/47.89968D0)
      IF(ALT .LT. 50.D0) B = (2.52D0*ALT - 90.5D0)*SCALE
      IF(ALT .GE. 50.D0) B = (0.2284D0*ALT + 21.775D0)*SCALE
      PABS = C + B
      TEMPK = 273.2D0 - 56.5D0
      GO TO 999
C
C  COMPUTE ALTITUDE FROM PRESSURE
C  CHECK FOR TROPOPAUSE
C
  200 IF(PABS .GE. .001D0) GOTO 205
      ALT = 170.D0
      GOTO 999
  205 IF(PABS .LT. 0.2236D0) GO TO 250
C
C   MODEL FOR BELOW TROPOPAUSE
C
      ALTOLD = TO/TC*(1.D0 - PABS**(1/X))/1000.D0*3.281D0
  210 ARG = (2.15287D0*ALTOLD - 8.16554D0)*SCALE
      ALT = TO/1.9812D0*(1.D0 - (PABS - ARG)**(1.D0/X))
      TEMPK = TO - TC*ALT
      IF(ABS(ALT - ALTOLD) .LT. 1.D-4) GO TO 999
      ALTOLD = (ALT + ALTOLD)/2.D0
      GO TO 210
C
C   MODEL FOR ABOVE TROPOPAUSE
C
C     CHECK FOR 50000 FEET OR ABOVE.
C
  250 ALTOLD = (DLOG10(0.2236D0/PABS)*14.6D0 + 11.D0)*3.281D0
  260 B = (2.52D0*ALTOLD - 90.5D0)*SCALE
```

```
      IF(PABS .LT. 0.11511600) B = (0.2284D0*ALTOLD+21.775D0)*SCALE
      ALT = 36.088900 - 47.89968D0*DLOG10((PABS - B)/0.224D0)
      TEMPK = 273.2D0 - 56.5D0
      IF(ABS(ALT - ALTOLD) .LT. 1.D-4) GO TO 999
      ALTOLD = (ALT + ALTOLD)/2.D0
      GO TO 260
  999 ALT = ALT*1000.D0
      RETURN
      END
      SUBROUTINE MILCAB(PABACF,PABCAB,PMAX,IUNITS)
C
C  DOUBLE PRECISION VERSION
C
C
C       U.S MILSPEC CABIN PRESSURIZATION SCHEDULE AS DETAILED
C       IN MIL-E-38453A (USAF) 2 DEC 1971.
C
C    GIVEN ABSOLUTE PRESSURE OF AIRCRAFT MILCAB RETURNS
C    THE CABIN ABSOLUTE PRESSURE OR VICE VERSA.
C      TO COMPUTE CABIN FROM ACFT SET INVERT .GE. 0.
C
C   PRESSURE UNITS SET BY IUNITS
C     SET IUNITS = 0 FOR PSIA
C     SET IUNITS = 1 FOR MM HG
C
C   PMAX = THE MAXIMUM DIFFERENTIAL PRESSURE PRODUCED BY THE ECS
C
C
C
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      INTEGER IUNITS
      CPMM = 51.72D0
      IF(IUNITS .EQ. 1) PABACF = PABACF/CPMM
      IF(PABACF .GT. 10.92D0) GO TO 110
      GO TO 120
  110 PABCAB = PABACF
      GO TO 999
  120 IF(PABACF .LE. 10.917D0 - PMAX) GO TO 130
      PABCAB = 10.917D0
      GO TO 999
  130 PABCAB = PABACF + PMAX
  999 IF(IUNITS .NE. 1) RETURN
      PABCAB = PABCAB*CPMM
      PABACF = PABACF*CPMM
      RETURN
      END
      DOUBLE PRECISION FUNCTION PBFORA(PCH)
C
C  COMPUTES PRESSURE BREATHING FOR ALTITUDE AS A FUNCTION OF AMBIENT PRESS
C
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      IF(PCH .GT. 0.2D0) THEN
      PBFORA = 0.0D0
      ELSE
      PBFORA = DMAX1(0.04D0,0.094D0-0.49D0*PCH)
      END IF
      RETURN
      END
      DOUBLE PRECISION FUNCTION DVL(DPLUNG)
C
C   COMPUTES FUNCTIONAL RESIDUAL VOLUME AS FUNCTION
C   OF DIFFERENTIAL BREATHING PRESSURE
C
      DOUBLE PRECISION DPLUNG
      DVL =  2.2502D-7*DPLUNG
```

```
      RETURN
      END
      SUBROUTINE PULFCN(TIME,PPBALT,VDOTE,QPEAK,QRESP,VLUNG)
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      DOUBLE PRECISION  KG, KP1, KP1OK, KOKM1, KOKP1
      COMMON/T/         T, XT,     NSTOP,      NORUN
      COMMON /R/ PI, P0, V0, T0, RUNIV,  ! Pi = 3.1412..., Std P,V,T& R
     *          PA2MM,
     *          KG, KP1, KOKM1, KP1OK, KOKP1,! K, K/(K-1), (K+1)/K, 1/K, 2/K, etc
     *          TWOOK, THKOK, TWOOKP1,  ! Adiabatic constant and
     *          TWOOKM1, ONEOK,         ! derived constants
     *          CD,                     ! Orifice Discharge Coefficient
     *          TBODY, TAMB, PAMB,            ! Temperatures - Body & Ambient
     *          GMW, GMWS, GASK, GASKS, ! Gas parameters MW & Specif Heat ratios
     *          QAM, OMEGA, TINSP, FRM,       ! Breathing flow parameters
     *          TR, TPAUSE, RR, VTIDAL, ! Breathing flow parameters
     *          VFRC0, FRC, PPWBODY,   ! Initial, Current FRC, ppH2O Body
     *          AM(2), AMI(3), AME(3),  ! Airway parameters
     *          VAO, VMAO, CINERT, DPAW,! Airway parameters
     *          AWI, DVL,
     *          PREGO, PREF, Gdot1,     ! Regulator Outlet Pressure
     *          VDSMSK, AMIV, AMEV,     ! Mask parameters
     *          QIV, QEV,               ! Mask valve flow indicators
     *          VOL1, VOL2, VOL3, VOL4, ! Volumes
     *          AWN, AWO, AWH, AWAr,    ! Atomic wts
     *          GMN2, GMO2, GMCO2,      ! Molecular wts
     *          GMH2O, GMWAIR,          !    "     "
     *          FINSP(4), FEXP(4)       !Gas Fractions, O2,N2+Ar,CO2,H2O
C
      QOFT(T,QM,W) = QM*DSIN(W*T)
      VLOFT(T,QM,W) = QM/W*(1.0D0-DCOS(W*T))
      VFRC(PB) = 2.5D0 + 0.1D0*PB*760.D0
      TWOPI = 2.D0*PI
C
C   PEAK FLOW = VDOTE*PI FOR A HALF WAVE RECTIFIED SINUSOIDAL
C   DEMAND FLOW PATTERN.
C
      QPEAK = VDOTE*PI
C
C   USE THE CURVE FIT VTID TO ESTIMATE TIDAL VOLUME FROM AVERAGE FLOW.
C   THEN DIVIDE BY VE TO GET TAV, THE AVERAGE TIME PER BREATHING CYCLE.
C
      CALL VTIDL(VDOTE,VTIDAL)
      TAV = VTIDAL/VDOTE
C
C   COMPUTE RESPIRATION RATE FROM  RESP RATE (BREATHS/TIME) = 1/TAV
C
      RR = 1.D0/TAV
C
C   SCALE TIME FROM MINUTES TO SECONDS/CYCLE
C
      TAV = TAV*60.D0
C
C   COMPUTE THE AVERAGE RADIAN BREATHING FREQUENCY
C
      OMEGA = TWOPI/TAV
      QRESP = QOFT(TIME,QPEAK,OMEGA)
      VLUNG = VFRC(PPBALT) + VLOFT(TIME,QPEAK,OMEGA)/60.D0
      RETURN
      END
C...
C...
C...
      SUBROUTINE VTIDL(VE,VT)
C   DOUBLE PRECISION VERSION
```

Program MASKSUBS.FOR

```
C
C  ESTIMATE TIDAL VOLUME FROM PULMONARY VENTILATION (V DOT E) ACCORDING TO
C  A THE MODEL OF HEY ET AL, RESPIRATION PHYSIOLOGY (1966) V1,191-205.
C
C
C   USE HEY'S MEAN MODEL PARAMETERS FOR "M" AND "K"
C   (MBAR = 28 +/- 2 [1/MIN] AND KBAR = 0.31 +/- 0.08 [LITERS])
C
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      DOUBLE PRECISION MBAR,KBAR
      MBAR = 28.D0
      KBAR = 0.31D0
      VT = VE/MBAR + KBAR
      RETURN
      END
C...
C...
C...
      SUBROUTINE FIOX(PAO2,PACO2,R,PAB,FIO2,FIO2MX,PINSP,PAOXY)
C
C   DOUBLE PRECISION VERSION
C
C   FIOX EMPLOYS THE ALVEOLAR AIR EQUATION FOR STEADY STATE EXCHANGE
C   TO CALCULATE THE MINIMUM FRACTION INSPIRED O2 (FIO2) REQUIRED
C   TO MAINTAIN PAO2 GIVEN PACO2, R, AND ABSOLUTE PRESSURE(PAB).
C   PRESSURES MUST BE IN MM HG. IF PRESSURE BREATHING IS REQUIRED
C   FIO2 IS SET TO 1.0 AND THE REQUIRED SAFETY PRESSURE IS RETURNED IN PINSP
C   IN MM HG(GAUGE).  THE ESTIMATED ALVEOLAR OXYGEN PARTIAL PRESSURE WITHOUT
C   SAFETY PRESSURE IS RETURNED IN PAOXY.
C
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      OMROR = (1.D0 - R)/R
      A = PAO2 + PACO2/R
      B = PACO2*OMROR + PAB - 47.D0
      FIO2 = A/B
      PAOXY = PAO2
      PINSP = 0.0D0
      IF(FIO2 .GE. 0.D0 .AND. FIO2 .LE. FIO2MX) RETURN
      IF(FIO2 .LT. 0.D0) GO TO 999
      PC = 47.D0 - PACO2*OMROR + A/FIO2MX
      PINSP = PC - PAB
      FIO2 = FIO2MX
      BNEW = PACO2*OMROR + PC - 47.D0
      PAOXY = FIO2MX*BNEW - PACO2/R
      RETURN
  999 WRITE(*,2) FIO2
      FIO2 = 0.
    2 FORMAT(5X,'***ERROR IN FIOX, FIO2 = ',1PE12.3)
      RETURN
      END
C...
C...
C...
      SUBROUTINE RDO2(PAO2MN,FRIO2,FIO2MX,PABACF,PABCAB,PSAFTI,
     *PSAFTF,PACO2,R)
C
C   DOUBLE PRECISION VERSION
C
C   RDO2 COMPUTES MINIMUM FRACTION INSP O2 REQUIRED TO MAINTAIN
C   30MM HG PAO2 FOLLOWING DECOMPRESSION FROM PABCAB TO PABACF.
C   CORRECTION FOR SAFETY PRESSURE IS MADE. REQUIRES
C   SUBROUTINE FIO2(PAO2,PACO2,R,PABCAB,FIO2,FIO2MX,PINSP,PAOXY)
C   PRESSURES MUST BE SUPPLIED IN MM HG.  PABACF, PABCAB & PACO2
C   ARE ABSOLUTE PRESSURES, PSAFTI IS GAUGE PRESSURE(MASK LESS CABIN).
C   BEFORE DECOMPRESSION PSAFTF IS SAFETY PRESSURE AFTER DECOMPRESSION.
```

Program MASKSUBS.FOR

```
C
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      PLI = PABCAB + PSAFTI - 47.D0  ! mmHg
      PLF = PABACF + PSAFTF - 47.D0  ! mmHg
      PAO2 = PAO2MN*PLI/PLF
      CALL FIOX(PAO2,PACO2,R,PABCAB+PSAFTI,FRIO2,FIO2MX,PINSP,PAOXY)
      RETURN
      END
C...
C...
C...
      SUBROUTINE ALVPO2(FIO2,PACO2,R,PAB,PINSP,PAOXY,EQVALT)
C
C    DOUBLE PRECISION VERSION
C
C   SUBROUTINE ALVPO2 EMPLOYS THE ALVEOLAR AIR EQUATION TO ESTIMATE
C   THE STEADY STATE ALVEOLAR OXYGEN PARTIAL PRESSURE (PAOXY), THE
C   CORRESPONDING ALVEOLAR CARBON DIOXIDE PARTIAL PRESSURE (PACO2)
C   AND RESPIRATORY EXCHANGE RATIO, (R) GIVEN THE FRACTION INSPIRED
C   OXYGEN (FIO2), THE ABSOLUTE AMBIENT PRESSURE (PAB) AND THE LEVEL OF
C   DIFFERENTIAL PRESSURE BREATHING (PINSP).  THE SUBROUTINE ALSO
C   CALCULATES THE ALTITUDE (EQVALT) AT WHICH THE EQUIVALENT PAOXY IS
C   PRODUCED IF AIR IS BREATHED.
C
C
C
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
  100 SAVPO2 = PAOXY
      SAVCO2 = PACO2
      OMROR = (1.0D0 - R)/R
      PABRSP = PAB + PINSP
      PAOXY = FIO2*(PACO2*OMROR + PABRSP - 47.D0) - PACO2/R
      IF(PAOXY .LE. 0.0D0) THEN
      PAOXY = FIO2*(PABRSP - 47) - PACO2
      ELSE
      PACO2 = 1.D0/(0.0207D0 + 0.474D0/PAOXY)
      END IF
      IF(PAOXY .LE. 25.D0) THEN
         R = 1.05D0
      ELSE
         R = 5.D0*DLOG(PACO2)-0.7843D0*ALOG(PACO2)**2.D0-6.9266D0
      END IF
  500 IF(ABS(SAVPO2 - PAOXY) .LE. 0.01D0) THEN
       GO TO 1000
      ELSE
       PAOXY = (SAVPO2+PAOXY)/2.D0
       PACO2 = (SAVCO2+PACO2)/2.D0
       GO TO 100
      END IF
 1000 PABRSP = (PAOXY + PACO2/R)/0.2095D0 - PACO2*OMROR + 47.D0
      PABRSP = PABRSP/759.9D0
      CALL DICAN(-1,EQVALT,PABRSP,TEMPK)
      RETURN
      END
```

Program RKF45.FOR

```
*DECK RKF45
      SUBROUTINE RKF45(F,NEQN,Y,T,TOUT,RELERR,ABSERR,IFLAG,WORK,IWORK)
C
C     FEHLBERG FOURTH-FIFTH ORDER RUNGE-KUTTA METHOD
C
C     WRITTEN BY H.A.WATTS AND L.F.SHAMPINE
C                   SANDIA LABORATORIES
C                  ALBUQUERQUE,NEW MEXICO
C
C    RKF45 IS PRIMARILY DESIGNED TO SOLVE NON-STIFF AND MILDLY STIFF
C    DIFFERENTIAL EQUATIONS WHEN DERIVATIVE EVALUATIONS ARE INEXPENSIVE.
C    RKF45 SHOULD GENERALLY NOT BE USED WHEN THE USER IS DEMANDING
C    HIGH ACCURACY.
C
C ABSTRACT
C
C    SUBROUTINE  RKF45  INTEGRATES A SYSTEM OF NEQN FIRST ORDER
C    ORDINARY DIFFERENTIAL EQUATIONS OF THE FORM
C             DY(I)/DT = F(T,Y(1),Y(2),...,Y(NEQN))
C              WHERE THE Y(I) ARE GIVEN AT T .
C    TYPICALLY THE SUBROUTINE IS USED TO INTEGRATE FROM T TO TOUT BUT IT
C    CAN BE USED AS A ONE-STEP INTEGRATOR TO ADVANCE THE SOLUTION A
C    SINGLE STEP IN THE DIRECTION OF TOUT.  ON RETURN THE PARAMETERS IN
C    THE CALL LIST ARE SET FOR CONTINUING THE INTEGRATION. THE USER HAS
C    ONLY TO CALL RKF45 AGAIN (AND PERHAPS DEFINE A NEW VALUE FOR TOUT).
C    ACTUALLY, RKF45 IS AN INTERFACING ROUTINE WHICH CALLS SUBROUTINE
C    RKFS FOR THE SOLUTION.  RKFS IN TURN CALLS SUBROUTINE  FEHL WHICH
C    COMPUTES AN APPROXIMATE SOLUTION OVER ONE STEP.
C
C    RKF45  USES THE RUNGE-KUTTA-FEHLBERG (4,5)  METHOD DESCRIBED
C    IN THE REFERENCE
C    E.FEHLBERG , LOW-ORDER CLASSICAL RUNGE-KUTTA FORMULAS WITH STEPSIZE
C                 CONTROL , NASA TR R-315
C
C    THE PERFORMANCE OF RKF45 IS ILLUSTRATED IN THE REFERENCE
C    L.F.SHAMPINE,H.A.WATTS,S.DAVENPORT, SOLVING NON-STIFF ORDINARY
C                 DIFFERENTIAL EQUATIONS-THE STATE OF THE ART ,
C                 SANDIA LABORATORIES REPORT SAND75-0182 ,
C                 TO APPEAR IN SIAM REVIEW.
C
C
C    THE PARAMETERS REPRESENT-
C      F -- SUBROUTINE F(T,Y,YP) TO EVALUATE DERIVATIVES YP(I)=DY(I)/DT
C      NEQN -- NUMBER OF EQUATIONS TO BE INTEGRATED
C      Y(*) -- SOLUTION VECTOR AT T
C      T -- INDEPENDENT VARIABLE
C      TOUT -- OUTPUT POINT AT WHICH SOLUTION IS DESIRED
C      RELERR,ABSERR -- RELATIVE AND ABSOLUTE ERROR TOLERANCES FOR LOCAL
C            ERROR TEST. AT EACH STEP THE CODE REQUIRES THAT
C                 ABS(LOCAL ERROR) .LE. RELERR*ABS(Y) + ABSERR
C            FOR EACH COMPONENT OF THE LOCAL ERROR AND SOLUTION VECTORS
C      IFLAG -- INDICATOR FOR STATUS OF INTEGRATION
C      WORK(*) -- ARRAY TO HOLD INFORMATION INTERNAL TO RKF45 WHICH IS
C            NECESSARY FOR SUBSEQUENT CALLS. MUST BE DIMENSIONED
C            AT LEAST  3+6*NEQN
C      IWORK(*) -- INTEGER ARRAY USED TO HOLD INFORMATION INTERNAL TO
C            RKF45 WHICH IS NECESSARY FOR SUBSEQUENT CALLS. MUST BE
C            DIMENSIONED AT LEAST  5
C
C
C  FIRST CALL TO RKF45
C
C    THE USER MUST PROVIDE STORAGE IN HIS CALLING PROGRAM FOR THE ARRAYS
C    IN THE CALL LIST  -      Y(NEQN) , WORK(3+6*NEQN) , IWORK(5)  ,
C    DECLARE F IN AN EXTERNAL STATEMENT, SUPPLY SUBROUTINE F(T,Y,YP) AND
```

Program RKF45.FOR

```
C    INITIALIZE THE FOLLOWING PARAMETERS-
C
C      NEQN -- NUMBER OF EQUATIONS TO BE INTEGRATED.  (NEQN .GE. 1)
C      Y(*) -- VECTOR OF INITIAL CONDITIONS
C      T -- STARTING POINT OF INTEGRATION , MUST BE A VARIABLE
C      TOUT -- OUTPUT POINT AT WHICH SOLUTION IS DESIRED.
C            T=TOUT IS ALLOWED ON THE FIRST CALL ONLY, IN WHICH CASE
C            RKF45 RETURNS WITH IFLAG=2 IF CONTINUATION IS POSSIBLE.
C      RELERR,ABSERR -- RELATIVE AND ABSOLUTE LOCAL ERROR TOLERANCES
C            WHICH MUST BE NON-NEGATIVE. RELERR MUST BE A VARIABLE WHILE
C            ABSERR MAY BE A CONSTANT. THE CODE SHOULD NORMALLY NOT BE
C            USED WITH RELATIVE ERROR CONTROL SMALLER THAN ABOUT 1.E-8 .
C            TO AVOID LIMITING PRECISION DIFFICULTIES THE CODE REQUIRES
C            RELERR TO BE LARGER THAN AN INTERNALLY COMPUTED RELATIVE
C            ERROR PARAMETER WHICH IS MACHINE DEPENDENT. IN PARTICULAR,
C            PURE ABSOLUTE ERROR IS NOT PERMITTED. IF A SMALLER THAN
C            ALLOWABLE VALUE OF RELERR IS ATTEMPTED, RKF45 INCREASES
C            RELERR APPROPRIATELY AND RETURNS CONTROL TO THE USER BEFORE
C            CONTINUING THE INTEGRATION.
C      IFLAG -- +1,-1  INDICATOR TO INITIALIZE THE CODE FOR EACH NEW
C            PROBLEM. NORMAL INPUT IS +1. THE USER SHOULD SET IFLAG=-1
C            ONLY WHEN ONE-STEP INTEGRATOR CONTROL IS ESSENTIAL. IN THIS
C            CASE, RKF45 ATTEMPTS TO ADVANCE THE SOLUTION A SINGLE STEP
C            IN THE DIRECTION OF TOUT EACH TIME IT IS CALLED. SINCE THIS
C            MODE OF OPERATION RESULTS IN EXTRA COMPUTING OVERHEAD, IT
C            SHOULD BE AVOIDED UNLESS NEEDED.
C
C
C   OUTPUT FROM RKF45
C
C      Y(*) -- SOLUTION AT T
C      T -- LAST POINT REACHED IN INTEGRATION.
C      IFLAG = 2 -- INTEGRATION REACHED TOUT. INDICATES SUCCESSFUL RETURN
C                   AND IS THE NORMAL MODE FOR CONTINUING INTEGRATION.
C            =-2 -- A SINGLE SUCCESSFUL STEP IN THE DIRECTION OF TOUT
C                   HAS BEEN TAKEN. NORMAL MODE FOR CONTINUING
C                   INTEGRATION ONE STEP AT A TIME.
C            = 3 -- INTEGRATION WAS NOT COMPLETED BECAUSE RELATIVE ERROR
C                   TOLERANCE WAS TOO SMALL. RELERR HAS BEEN INCREASED
C                   APPROPRIATELY FOR CONTINUING.
C            = 4 -- INTEGRATION WAS NOT COMPLETED BECAUSE MORE THAN
C                   3000 DERIVATIVE EVALUATIONS WERE NEEDED. THIS
C                   IS APPROXIMATELY 500 STEPS.
C            = 5 -- INTEGRATION WAS NOT COMPLETED BECAUSE SOLUTION
C                   VANISHED MAKING A PURE RELATIVE ERROR TEST
C                   IMPOSSIBLE. MUST USE NON-ZERO ABSERR TO CONTINUE.
C                   USING THE ONE-STEP INTEGRATION MODE FOR ONE STEP
C                   IS A GOOD WAY TO PROCEED.
C            = 6 -- INTEGRATION WAS NOT COMPLETED BECAUSE REQUESTED
C                   ACCURACY COULD NOT BE ACHIEVED USING SMALLEST
C                   ALLOWABLE STEPSIZE. USER MUST INCREASE THE ERROR
C                   TOLERANCE BEFORE CONTINUED INTEGRATION CAN BE
C                   ATTEMPTED.
C            = 7 -- IT IS LIKELY THAT RKF45 IS INEFFICIENT FOR SOLVING
C                   THIS PROBLEM. TOO MUCH OUTPUT IS RESTRICTING THE
C                   NATURAL STEPSIZE CHOICE. USE THE ONE-STEP INTEGRATOR
C                   MODE.
C            = 8 -- INVALID INPUT PARAMETERS
C                   THIS INDICATOR OCCURS IF ANY OF THE FOLLOWING IS
C                   SATISFIED -   NEQN .LE. 0
C                                 T=TOUT  AND  IFLAG .NE. +1 OR -1
C                                 RELERR OR ABSERR .LT. 0.
C                                 IFLAG .EQ. 0  OR  .LT. -2  OR  .GT. 8
C      WORK(*),IWORK(*) -- INFORMATION WHICH IS USUALLY OF NO INTEREST
C                   TO THE USER BUT NECESSARY FOR SUBSEQUENT CALLS.
```

Program RKF45.FOR

```
C                   WORK(1),...,WORK(NEQN) CONTAIN THE FIRST DERIVATIVES
C                   OF THE SOLUTION VECTOR Y AT T. WORK(NEQN+1) CONTAINS
C                   THE STEPSIZE H TO BE ATTEMPTED ON THE NEXT STEP.
C                   IWORK(1) CONTAINS THE DERIVATIVE EVALUATION COUNTER.
C
C
C  SUBSEQUENT CALLS TO RKF45
C
C   SUBROUTINE RKF45 RETURNS WITH ALL INFORMATION NEEDED TO CONTINUE
C   THE INTEGRATION. IF THE INTEGRATION REACHED TOUT, THE USER NEED ONL
C   DEFINE A NEW TOUT AND CALL RKF45 AGAIN. IN THE ONE-STEP INTEGRATOR
C   MODE (IFLAG=-2) THE USER MUST KEEP IN MIND THAT EACH STEP TAKEN IS
C   IN THE DIRECTION OF THE CURRENT TOUT. UPON REACHING TOUT (INDICATED
C   BY CHANGING IFLAG TO 2),THE USER MUST THEN DEFINE A NEW TOUT AND
C   RESET IFLAG TO -2 TO CONTINUE IN THE ONE-STEP INTEGRATOR MODE.
C
C   IF THE INTEGRATION WAS NOT COMPLETED BUT THE USER STILL WANTS TO
C   CONTINUE (IFLAG=3,4 CASES), HE JUST CALLS RKF45 AGAIN. WITH IFLAG=3
C   THE RELERR PARAMETER HAS BEEN ADJUSTED APPROPRIATELY FOR CONTINUING
C   THE INTEGRATION. IN THE CASE OF IFLAG=4 THE FUNCTION COUNTER WILL
C   BE RESET TO 0 AND ANOTHER 3000 FUNCTION EVALUATIONS ARE ALLOWED.
C
C   HOWEVER,IN THE CASE IFLAG=5, THE USER MUST FIRST ALTER THE ERROR
C   CRITERION TO USE A POSITIVE VALUE OF ABSERR BEFORE INTEGRATION CAN
C   PROCEED. IF HE DOES NOT,EXECUTION IS TERMINATED.
C
C   ALSO,IN THE CASE IFLAG=6, IT IS NECESSARY FOR THE USER TO RESET
C   IFLAG TO 2 (OR -2 WHEN THE ONE-STEP INTEGRATION MODE IS BEING USED)
C   AS WELL AS INCREASING EITHER ABSERR,RELERR OR BOTH BEFORE THE
C   INTEGRATION CAN BE CONTINUED. IF THIS IS NOT DONE, EXECUTION WILL
C   BE TERMINATED. THE OCCURRENCE OF IFLAG=6 INDICATES A TROUBLE SPOT
C   (SOLUTION IS CHANGING RAPIDLY,SINGULARITY MAY BE PRESENT) AND IT
C   OFTEN IS INADVISABLE TO CONTINUE.
C
C   IF IFLAG=7 IS ENCOUNTERED, THE USER SHOULD USE THE ONE-STEP
C   INTEGRATION MODE WITH THE STEPSIZE DETERMINED BY THE CODE OR
C   CONSIDER SWITCHING TO THE ADAMS CODES DE/STEP,INTRP. IF THE USER
C   INSISTS UPON CONTINUING THE INTEGRATION WITH RKF45, HE MUST RESET
C   IFLAG TO 2 BEFORE CALLING RKF45 AGAIN. OTHERWISE,EXECUTION WILL BE
C   TERMINATED.
C
C   IF IFLAG=8 IS OBTAINED, INTEGRATION CAN NOT BE CONTINUED UNLESS
C   THE INVALID INPUT PARAMETERS ARE CORRECTED.
C
C   IT SHOULD BE NOTED THAT THE ARRAYS WORK,IWORK CONTAIN INFORMATION
C   REQUIRED FOR SUBSEQUENT INTEGRATION. ACCORDINGLY, WORK AND IWORK
C   SHOULD NOT BE ALTERED.
C
C  WORK - Dimensioned for max NEQN = 500
C
      INTEGER NEQN,IFLAG,IWORK(5)
      DOUBLE PRECISION Y(NEQN),T,TOUT,RELERR,ABSERR,WORK(11000)
C     IF COMPILER CHECKS SUBSCRIPTS, CHANGE WORK(1) TO WORK(3+6*NEQN)
C
      EXTERNAL F
C
      INTEGER K1,K2,K3,K4,K5,K6,K1M
C
C
C     COMPUTE INDICES FOR THE SPLITTING OF THE WORK ARRAY
C
      K1M=NEQN+1
      K1=K1M+1
      K2=K1+NEQN
      K3=K2+NEQN
```

Program RKF45.FOR

```
      K4=K3+NEQN
      K5=K4+NEQN
      K6=K5+NEQN
C
C     THIS INTERFACING ROUTINE MERELY RELIEVES THE USER OF A LONG
C     CALLING LIST VIA THE SPLITTING APART OF TWO WORKING STORAGE
C     ARRAYS. IF THIS IS NOT COMPATIBLE WITH THE USERS COMPILER,
C     HE MUST USE RKFS DIRECTLY.
C
      CALL RKFS(F,NEQN,Y,T,TOUT,RELERR,ABSERR,IFLAG,WORK(1),WORK(K1M),
     1          WORK(K1),WORK(K2),WORK(K3),WORK(K4),WORK(K5),WORK(K6),
     2          WORK(K6+1),IWORK(1),IWORK(2),IWORK(3),IWORK(4),IWORK(5))
C
      RETURN
      END
      SUBROUTINE RKFS(F,NEQN,Y,T,TOUT,RELERR,ABSERR,IFLAG,YP,H,F1,F2,F3,
     1                F4,F5,SAVRE,SAVAE,NFE,KOP,INIT,JFLAG,KFLAG)
C
C     FEHLBERG FOURTH-FIFTH ORDER RUNGE-KUTTA METHOD
C
C
C     RKFS INTEGRATES A SYSTEM OF FIRST ORDER ORDINARY DIFFERENTIAL
C     EQUATIONS AS DESCRIBED IN THE COMMENTS FOR RKF45 .
C     THE ARRAYS YP,F1,F2,F3,F4,AND F5 (OF DIMENSION AT LEAST NEQN) AND
C     THE VARIABLES H,SAVRE,SAVAE,NFE,KOP,INIT,JFLAG,AND KFLAG ARE USED
C     INTERNALLY BY THE CODE AND APPEAR IN THE CALL LIST TO ELIMINATE
C     LOCAL RETENTION OF VARIABLES BETWEEN CALLS. ACCORDINGLY, THEY
C     SHOULD NOT BE ALTERED. ITEMS OF POSSIBLE INTEREST ARE
C         YP - DERIVATIVE OF SOLUTION VECTOR AT T
C         H  - AN APPROPRIATE STEPSIZE TO BE USED FOR THE NEXT STEP
C         NFE- COUNTER ON THE NUMBER OF DERIVATIVE FUNCTION EVALUATIONS
C
C
      LOGICAL HFAILD,OUTPUT
C
      INTEGER  NEQN,IFLAG,NFE,KOP,INIT,JFLAG,KFLAG
      DOUBLE PRECISION  Y(NEQN),T,TOUT,RELERR,ABSERR,H,YP(NEQN),
     1  F1(NEQN),F2(NEQN),F3(NEQN),F4(NEQN),F5(NEQN),SAVRE,
     2  SAVAE
C
      EXTERNAL F
C
      DOUBLE PRECISION  A,AE,DT,EE,EEOET,ESTTOL,ET,HMIN,REMIN,RER,S,
     1  SCALE,TOL,TOLN,U26,EPSP1,EPS,YPK
C
      INTEGER  K,MAXNFE,MFLAG
C
      DOUBLE PRECISION  DABS,DMAX1,DMIN1,DSIGN
C
C     REMIN IS THE MINIMUM ACCEPTABLE VALUE OF RELERR.  ATTEMPTS
C     TO OBTAIN HIGHER ACCURACY WITH THIS SUBROUTINE ARE USUALLY
C     VERY EXPENSIVE AND OFTEN UNSUCCESSFUL.
C
      DATA REMIN/1.0D-12/
C
C
C     THE EXPENSE IS CONTROLLED BY RESTRICTING THE NUMBER
C     OF FUNCTION EVALUATIONS TO BE APPROXIMATELY MAXNFE.
C     AS SET, THIS CORRESPONDS TO ABOUT 500 STEPS.
C
      DATA MAXNFE/2000000/
C
C
C     CHECK INPUT PARAMETERS
C
```

Program RKF45.FOR

```
C
      IF (NEQN .LT. 1) GO TO 10
      IF ((RELERR .LT. 0.0D0)  .OR.  (ABSERR .LT. 0.0D0)) GO TO 10
      MFLAG=IABS(IFLAG)
      IF ((MFLAG .EQ. 0) .OR. (MFLAG .GT. 8)) GO TO 10
      IF (MFLAG .NE. 1) GO TO 20
C
C     FIRST CALL, COMPUTE MACHINE EPSILON
C
      EPS = 1.0D0
    5 EPS = EPS/2.0D0
      EPSP1 = EPS + 1.0D0
      IF (EPSP1 .GT. 1.0D0) GO TO 5
      U26 = 26.0D0*EPS
      GO TO 50
C
C     INVALID INPUT
   10 IFLAG=8
      RETURN
C
C     CHECK CONTINUATION POSSIBILITIES
C
   20 IF ((T .EQ. TOUT) .AND. (KFLAG .NE. 3)) GO TO 10
      IF (MFLAG .NE. 2) GO TO 25
C
C     IFLAG = +2 OR -2
      IF ((KFLAG .EQ. 3) .OR. (INIT .EQ. 0)) GO TO 45
      IF (KFLAG .EQ. 4) GO TO 40
      IF ((KFLAG .EQ. 5)  .AND.  (ABSERR .EQ. 0.0D0)) GO TO 30
      IF ((KFLAG .EQ. 6)  .AND.  (RELERR .LE. SAVRE)  .AND.
     1    (ABSERR .LE. SAVAE)) GO TO 30
      GO TO 50
C
C     IFLAG = 3,4,5,6,7 OR 8
   25 IF (IFLAG .EQ. 3) GO TO 45
      IF (IFLAG .EQ. 4) GO TO 40
      IF ((IFLAG .EQ. 5) .AND. (ABSERR .GT. 0.0D0)) GO TO 45
C
C     INTEGRATION CANNOT BE CONTINUED SINCE USER DID NOT RESPOND TO
C     THE INSTRUCTIONS PERTAINING TO IFLAG=5,6,7 OR 8
   30 STOP
C
C     RESET FUNCTION EVALUATION COUNTER
   40 NFE=0
      IF (MFLAG .EQ. 2) GO TO 50
C
C     RESET FLAG VALUE FROM PREVIOUS CALL
   45 IFLAG=JFLAG
      IF (KFLAG .EQ. 3) MFLAG=IABS(IFLAG)
C
C     SAVE INPUT IFLAG AND SET CONTINUATION FLAG VALUE FOR SUBSEQUENT
C     INPUT CHECKING
   50 JFLAG=IFLAG
      KFLAG=0
C
C     SAVE RELERR AND ABSERR FOR CHECKING INPUT ON SUBSEQUENT CALLS
      SAVRE=RELERR
      SAVAE=ABSERR
C
C     RESTRICT RELATIVE ERROR TOLERANCE TO BE AT LEAST AS LARGE AS
C     2*EPS+REMIN TO AVOID LIMITING PRECISION DIFFICULTIES ARISING
C     FROM IMPOSSIBLE ACCURACY REQUESTS
C
      RER=2.0D0*EPS+REMIN
      IF (RELERR .GE. RER) GO TO 55
```

Program RKF45.FOR

```
C
C     RELATIVE ERROR TOLERANCE TOO SMALL
      RELERR=RER
      IFLAG=3
      KFLAG=3
      RETURN
C
   55 DT=TOUT-T
C
      IF (MFLAG .EQ. 1) GO TO 60
      IF (INIT .EQ. 0) GO TO 65
      GO TO 80
C
C     INITIALIZATION --
C                         SET INITIALIZATION COMPLETION INDICATOR,INIT
C                         SET INDICATOR FOR TOO MANY OUTPUT POINTS,KOP
C                         EVALUATE INITIAL DERIVATIVES
C                         SET COUNTER FOR FUNCTION EVALUATIONS,NFE
C                         ESTIMATE STARTING STEPSIZE
C
   60 INIT=0
      KOP=0
C
      A=T
      CALL F(A,Y,YP)
      NFE=1
      IF (T .NE. TOUT) GO TO 65
      IFLAG=2
      RETURN
C
C
   65 INIT=1
      H=DABS(DT)
      TOLN=0.
      DO 70 K=1,NEQN
        TOL=RELERR*DABS(Y(K))+ABSERR
        IF (TOL .LE. 0.) GO TO 70
        TOLN=TOL
        YPK=DABS(YP(K))
        IF (YPK*H**5 .GT. TOL) H=(TOL/YPK)**0.2D0
   70 CONTINUE
      IF (TOLN .LE. 0.0D0) H=0.0D0
      H=DMAX1(H,U26*DMAX1(DABS(T),DABS(DT)))
      JFLAG=ISIGN(2,IFLAG)
C
C
C     SET STEPSIZE FOR INTEGRATION IN THE DIRECTION FROM T TO TOUT
C
   80 H=DSIGN(H,DT)
C
C     TEST TO SEE IF RKF45 IS BEING SEVERELY IMPACTED BY TOO MANY
C     OUTPUT POINTS
C
      IF (DABS(H) .GE. 2.0D0*DABS(DT)) KOP=KOP+1
      IF (KOP .NE. 100) GO TO 85
C
C     UNNECESSARY FREQUENCY OF OUTPUT
      KOP=0
      IFLAG=7
      RETURN
C
   85 IF (DABS(DT) .GT. U26*DABS(T)) GO TO 95
C
C     IF TOO CLOSE TO OUTPUT POINT,EXTRAPOLATE AND RETURN
C
```

```
      DO 90 K=1,NEQN
   90   Y(K)=Y(K)+DT*YP(K)
      A=TOUT
      CALL F(A,Y,YP)
      NFE=NFE+1
      GO TO 300
C
C
C     INITIALIZE OUTPUT POINT INDICATOR
C
   95 OUTPUT= .FALSE.
C
C     TO AVOID PREMATURE UNDERFLOW IN THE ERROR TOLERANCE FUNCTION,
C     SCALE THE ERROR TOLERANCES
C
      SCALE=2.0D0/RELERR
      AE=SCALE*ABSERR
C
C
C     STEP BY STEP INTEGRATION
C
  100 HFAILD= .FALSE.
C
C     SET SMALLEST ALLOWABLE STEPSIZE
C
      HMIN=U26*DABS(T)
C
C     ADJUST STEPSIZE IF NECESSARY TO HIT THE OUTPUT POINT.
C     LOOK AHEAD TWO STEPS TO AVOID DRASTIC CHANGES IN THE STEPSIZE AND
C     THUS LESSEN THE IMPACT OF OUTPUT POINTS ON THE CODE.
C
      DT=TOUT-T
      IF (DABS(DT) .GE. 2.0D0*DABS(H)) GO TO 200
      IF (DABS(DT) .GT. DABS(H)) GO TO 150
C
C     THE NEXT SUCCESSFUL STEP WILL COMPLETE THE INTEGRATION TO THE
C     OUTPUT POINT
C
      OUTPUT= .TRUE.
      H=DT
      GO TO 200
C
  150 H=0.5D0*DT
C
C
C
C     CORE INTEGRATOR FOR TAKING A SINGLE STEP
C
C     THE TOLERANCES HAVE BEEN SCALED TO AVOID PREMATURE UNDERFLOW IN
C     COMPUTING THE ERROR TOLERANCE FUNCTION ET.
C     TO AVOID PROBLEMS WITH ZERO CROSSINGS,RELATIVE ERROR IS MEASURED
C     USING THE AVERAGE OF THE MAGNITUDES OF THE SOLUTION AT THE
C     BEGINNING AND END OF A STEP.
C     THE ERROR ESTIMATE FORMULA HAS BEEN GROUPED TO CONTROL LOSS OF
C     SIGNIFICANCE.
C     TO DISTINGUISH THE VARIOUS ARGUMENTS, H IS NOT PERMITTED
C     TO BECOME SMALLER THAN 26 UNITS OF ROUNDOFF IN T.
C     PRACTICAL LIMITS ON THE CHANGE IN THE STEPSIZE ARE ENFORCED TO
C     SMOOTH THE STEPSIZE SELECTION PROCESS AND TO AVOID EXCESSIVE
C     CHATTERING ON PROBLEMS HAVING DISCONTINUITIES.
C     TO PREVENT UNNECESSARY FAILURES, THE CODE USES 9/10 THE STEPSIZE
C     IT ESTIMATES WILL SUCCEED.
C     AFTER A STEP FAILURE, THE STEPSIZE IS NOT ALLOWED TO INCREASE FOR
C     THE NEXT ATTEMPTED STEP. THIS MAKES THE CODE MORE EFFICIENT ON
C     PROBLEMS HAVING DISCONTINUITIES AND MORE EFFECTIVE IN GENERAL
```

```
C     SINCE LOCAL EXTRAPOLATION IS BEING USED AND EXTRA CAUTION SEEMS
C     WARRANTED.
C
C
C     TEST NUMBER OF DERIVATIVE FUNCTION EVALUATIONS.
C     IF OKAY,TRY TO ADVANCE THE INTEGRATION FROM T TO T+H
C
  200 IF (NFE .LE. MAXNFE) GO TO 220
C
C     TOO MUCH WORK
      IFLAG=4
      KFLAG=4
      RETURN
C
C     ADVANCE AN APPROXIMATE SOLUTION OVER ONE STEP OF LENGTH H
C
  220 CALL FEHL(F,NEQN,Y,T,H,YP,F1,F2,F3,F4,F5,F1)
      NFE=NFE+5
C
C     COMPUTE AND TEST ALLOWABLE TOLERANCES VERSUS LOCAL ERROR ESTIMATES
C     AND REMOVE SCALING OF TOLERANCES. NOTE THAT RELATIVE ERROR IS
C     MEASURED WITH RESPECT TO THE AVERAGE OF THE MAGNITUDES OF THE
C     SOLUTION AT THE BEGINNING AND END OF THE STEP.
C
      EEOET=0.0D0
      DO 250 K=1,NEQN
        ET=DABS(Y(K))+DABS(F1(K))+AE
        IF (ET .GT. 0.0D0) GO TO 240
C
C       INAPPROPRIATE ERROR TOLERANCE
        IFLAG=5
        RETURN
C
  240   EE=DABS((-2090.0D0*YP(K)+(21970.0D0*F3(K)-15048.0D0*F4(K)))+
     1                          (22528.0D0*F2(K)-27360.0D0*F5(K)))
  250   EEOET=DMAX1(EEOET,EE/ET)
C
      ESTTOL=DABS(H)*EEOET*SCALE/752400.0D0
C
      IF (ESTTOL .LE. 1.0D0) GO TO 260
C
C
C     UNSUCCESSFUL STEP
C                        REDUCE THE STEPSIZE , TRY AGAIN
C                        THE DECREASE IS LIMITED TO A FACTOR OF 1/10
C
      HFAILD= .TRUE.
      OUTPUT= .FALSE.
      S=0.1D0
      IF (ESTTOL .LT. 59049.0D0) S=0.9D0/ESTTOL**0.2D0
      H=S*H
      IF (DABS(H) .GT. HMIN) GO TO 200
C
C     REQUESTED ERROR UNATTAINABLE AT SMALLEST ALLOWABLE STEPSIZE
      IFLAG=6
      KFLAG=6
      RETURN
C
C
C     SUCCESSFUL STEP
C                        STORE SOLUTION AT T+H
C                        AND EVALUATE DERIVATIVES THERE
C
  260 T=T+H
      DO 270 K=1,NEQN
```

```
 270   Y(K)=F1(K)
      A=T
      CALL F(A,Y,YP)
      NFE=NFE+1
C
C
C                         CHOOSE NEXT STEPSIZE
C                         THE INCREASE IS LIMITED TO A FACTOR OF 5
C                         IF STEP FAILURE HAS JUST OCCURRED, NEXT
C                            STEPSIZE IS NOT ALLOWED TO INCREASE
C
      S=5.0D0
      IF (ESTTOL .GT. 1.889568D-4) S=0.9D0/ESTTOL**0.2D0
      IF (HFAILD) S=DMIN1(S,1.0D0)
      H=DSIGN(DMAX1(S*DABS(H),HMIN),H)
C
C     END OF CORE INTEGRATOR
C
C
C     SHOULD WE TAKE ANOTHER STEP
C
      IF (OUTPUT) GO TO 300
      IF (IFLAG .GT. 0) GO TO 100
C
C
C     INTEGRATION SUCCESSFULLY COMPLETED
C
C     ONE-STEP MODE
      IFLAG=-2
      RETURN
C
C     INTERVAL MODE
 300  T=TOUT
      IFLAG=2
      RETURN
C
      END
      SUBROUTINE FEHL(F,NEQN,Y,T,H,YP,F1,F2,F3,F4,F5,S)
C
C     FEHLBERG FOURTH-FIFTH ORDER RUNGE-KUTTA METHOD
C
C    FEHL INTEGRATES A SYSTEM OF NEQN FIRST ORDER
C    ORDINARY DIFFERENTIAL EQUATIONS OF THE FORM
C            DY(I)/DT=F(T,Y(1),---,Y(NEQN))
C    WHERE THE INITIAL VALUES Y(I) AND THE INITIAL DERIVATIVES
C    YP(I) ARE SPECIFIED AT THE STARTING POINT T. FEHL ADVANCES
C    THE SOLUTION OVER THE FIXED STEP H AND RETURNS
C    THE FIFTH ORDER (SIXTH ORDER ACCURATE LOCALLY) SOLUTION
C    APPROXIMATION AT T+H IN ARRAY S(I).
C    F1,---,F5 ARE ARRAYS OF DIMENSION NEQN WHICH ARE NEEDED
C    FOR INTERNAL STORAGE.
C    THE FORMULAS HAVE BEEN GROUPED TO CONTROL LOSS OF SIGNIFICANCE.
C    FEHL SHOULD BE CALLED WITH AN H NOT SMALLER THAN 13 UNITS OF
C    ROUNDOFF IN T SO THAT THE VARIOUS INDEPENDENT ARGUMENTS CAN BE
C    DISTINGUISHED.
C
C
      INTEGER  NEQN
      DOUBLE PRECISION  Y(NEQN),T,H,YP(NEQN),F1(NEQN),F2(NEQN),
     1  F3(NEQN),F4(NEQN),F5(NEQN),S(NEQN)
C
      DOUBLE PRECISION  CH
      INTEGER  K
C...
      EXTERNAL F
```

```
      CH=H/4.0D0
      DO 221 K=1,NEQN
  221   F5(K)=Y(K)+CH*YP(K)
      CALL F(T+CH,F5,F1)
C
      CH=3.0D0*H/32.0D0
      DO 222 K=1,NEQN
  222   F5(K)=Y(K)+CH*(YP(K)+3.0D0*F1(K))
      CALL F(T+3.0D0*H/8.0D0,F5,F2)
C
      CH=H/2197.0D0
      DO 223 K=1,NEQN
  223   F5(K)=Y(K)+CH*(1932.0D0*YP(K)+(7296.0D0*F2(K)-7200.0D0*F1(K)))
      CALL F(T+12.0D0*H/13.0D0,F5,F3)
C
      CH=H/4104.0D0
      DO 224 K=1,NEQN
  224   F5(K)=Y(K)+CH*((8341.0D0*YP(K)-845.0D0*F3(K))+
     1                          (29440.0D0*F2(K)-32832.0D0*F1(K)))
      CALL F(T+H,F5,F4)
C
      CH=H/20520.0D0
      DO 225 K=1,NEQN
  225   F1(K)=Y(K)+CH*((-6080.0D0*YP(K)+(9295.0D0*F3(K)-
     1         5643.0D0*F4(K)))+(41040.0D0*F1(K)-28352.0D0*F2(K)))
      CALL F(T+H/2.0D0,F1,F5)
C
C     COMPUTE APPROXIMATE SOLUTION AT T+H
C
      CH=H/7618050.0D0
      DO 230 K=1,NEQN
  230   S(K)=Y(K)+CH*((902880.0D0*YP(K)+(3855735.0D0*F3(K)-
     1        1371249.0D0*F4(K)))+(3953664.0D0*F2(K)+
     2        277020.0D0*F5(K)))
C
      RETURN
      END
```

APPENDIX C

Pulmonary Model Computer Programs

Appendix C - 1
Tissue Model - Part 1

PROGRAM TISSFCM

```
C  This program calculates the Fractional Capacity of lung parenchyma
C  for a given Pressure - Fractional Volume curve for a mechanical unit
c  by adding the effects of terminal bronchi closing and opening.

      DIMENSION CDF(100),FVtin(300),FVtex(300),SFVctb(100)

C COMMON STATEMENTS FOR SUBROUTINE IODELCS
      INTEGER INP, OUT, PLT
      INTEGER CRT, LUOUT, LUIN, LUPLT
      CHARACTER*1 BEEP
      CHARACTER*80 ITITLE
      CHARACTER*64 NAINP,NAOUT,NAPLT
      COMMON /DEVS/ LUTRM, LUIN, LUOUT, LUPLT
      COMMON /CURSOR/ BEEP
      COMMON /LABELS/ ITITLE, NAINP, NAOUT, NAPLT

C SETUP THE OUTPUT FILE
       CRT = 0
       CALL IODECLS(0,1,0)

C  CALCULATE THE EXHALATION P-FV CURVE FOR THE MECHANICAL UNIT

C  Define the normal distribution of closing pressures
C
      Pmean =  -0.8 !Pmean is the mean value of the closing pressures
      SIGMA = 0.5   !SIGMA is the Standard Deviation of the distribution
C     SIGMA = 0.8  for the inhalation distribution

C         Define the FV curve for the mechanical unit when
C                   all Terminal Bronchi are open

      Pzero =-5.0
      Phalf = 5.0
      ALPHA = .5/(Phalf-Pzero)
      BETA = (16*ALPHA**3)/27
C
C     FVmu(P) = ALPHA*(P-Pzero) - BETA*(P-Phalf)**3  for P>Phalf
C     FVmu(P) = ALPHA*(P-Pzero)    for P<Phalf
C
C  Divide the FVA curve into NMAX-1 intervals of DPe width, the Pressue Range
C  is 2.5*SIGMA on each side of the mean closing pressure (PM).
C
      Nmax = 21
      Prange = 2*(2.5*SIGMA)
      DPe = Prange/(Nmax-1)
C  The range is bounded by Pmaxe and Pmine.
C
      Pmaxe = Pmean + Prange/2
      Pmine = Pmean - Prange/2
C
C  Define a variable CDF at the end of each interval,
C  CDF = Cumulative Distribution Function of the closing pressures.
C

      WRITE(0,10)
10    FORMAT('                    ')
      WRITE(0,1)
1     FORMAT('       N       Pcc       CDF(N)')
2     FORMAT(5X,I4,4X,F6.2,4X,F6.3)
```

```
      Pcc = Pmaxe
      DO 100 N = Nmax,1,-1
      U = (Pcc - Pmean)/ SIGMA
      CDF(N) = (ERF(U) + 1)/2.0
      WRITE(0,2) N,Pcc,CDF(N)
100   Pcc = Pcc - DPe
C
      PAUSE'PAUSE AFTER THE CALCULATION OF THE CDF'
C
C  Calculate the FVtis in the region where Terminal Bronchi are closing
C
      WRITE(0,3)
 3    FORMAT('   PRESS    FTB OPEN    FVmu   N   FVtex    FVcpi   SFVc
     Xtb  FVotb')
 4    FORMAT(2X,F6.2,2X,F6.2,7X,F6.3,3X,I2,2X,F6.3,2X,F6.3,2X,F6.3,2X,
     xF6.3)
C
      DPhalfe = DPe/2          !Half a pressure increment DPe
      SFVctb(Nmax) = 0 !Summed Fractional Vol of Closed Terminal Bronchi
C
      Nmax1 = Nmax - 1
      DO 500 N = Nmax1,1,-1
      P = Pmine+(N-1)*DPe+DPhalfe       !P = average pressure in the nth interval
      FVmu = ALPHA*(P-Pzero)
      IF(P.GT.Phalf) THEN FVmu = FVmu - BETA*(P-Phalf)**3
C
      FVcpi = FVmu*(CDF(N+1) - CDF(N))!Fractional volume of gas that Closes
C                              during this Pressure Increment

      SFVctb(N) = SFVctb(N+1) + FVcpi!Summed Fractional Volume of all acini
C                             with Closed Terminal Bronchi
C
      FVotb = FVmu * CDF(N)           !Fractional Volume contribution of
C                             acini with Open Terminal Bronchi
C
      FVtex(N) = SFVctb(N) + FVotb     !Fractional Volume of the Tissue
C
500   WRITE(0,4) P,CDF(N),FVmu,N,FVtex(N),FVcpi,SFVctb(N),FVotb
      PAUSE' PAUSE AFTER OUTPUT OF THE FVtis curve where TB are closed'
C
C Complete FVtex curve for pressures where all terminal bronchi are open
C
      N= Nmax
700   P=Pmine +(N-1)*DPe +DPhalfe
      FVmu = ALPHA*(P-Pzero)
      IF(P.GT.Phalf) FVmu = FVmu -  BETA*(P-Phalf)**3
      FVtex(N) = FVmu
      IF(FVmu.GT.0.9999) GO TO 1000
      N = N+1
      GO TO 700
C
1000  Nmaxte = N      !#+1 of DPe increments in the FVtex curve
C
      J=0
      DO 1200 N=1,Nmaxte
      J = J+1
      IF(J.LT.10) GO TO 1100
      J=0
      PAUSE 'PAUSE DURING PRINT'
1100  P= Pmine + N*DPe - DPhalfe
1200  WRITE(0,9) N,P,FVtex(N)
 9    FORMAT(5X,I3,5X,F6.2,5X,F6.3)

C
C  Calculate the inhalation FVtis curve (FVtin) for a VC inhalation.
C  The inhalation curve for the mechanical unit has the same shape
C  as the exhalation curve
C  and all the P's are increased by an equal amount of hysteresis,
C  DPhys. The opening sequence is taken to be the opposite of the
```

```
C  closing sequence; the first terminal bronchi to close are the last
C  to open.

       SIGMA=0.8
       Pmean=Pmean+5.0
       DPhys=4.0
       Pzero=Pzero+DPhys
       Phalf=Phalf+DPhys

       Prange=2*(2.5*SIGMA)
       DPi=Prange/(Nmax-1)
       DPhalfi=DPi/2.0
       Pmaxi=Pmean+Prange/2
       Pmini=Pmean-Prange/2

       FVtin(1)=FVtex(1)

       WRITE(0,1450)
 1450  FORMAT('    PRESS    FTB OPEN     FVmu    N    FVtin     FVcpi    SFVc
     Xtb  FVotb')

       DO  1500 N=2,Nmax
       P = Pmini+(N-1)*DPi-DPhalfi   !P = average pressure in the nth interval
       FVmu = ALPHA*(P-Pzero)
       IF(P.GT.Phalf) THEN FVmu = FVmu - BETA*(P-Phalf)**3
       SFVotb=CDF(N)*FVmu          !Summed FV of all open TB
       FVtin(N)=SFVotb+SFVctb(N)   !summed FV aof all open and closed TB
 1500  WRITE(0,4) P,CDF(N),FVmu,N,FVtin(N),FVcpi,SFVctb(N),SFVotb
      PAUSE' Pause after output of FVtin curve where all TB have just
     x opened'
C
C Complete FVtin curve for pressures where all terminal bronchi are open
C
       N= Nmax
 1700  P=Pmini +(N-1)*DPi +DPhalfi
       FVmu = ALPHA*(P-Pzero)
       IF(P.GT.Phalf) FVmu = FVmu -  BETA*(P-Phalf)**3
       FVtin(N) = FVmu
       IF(FVmu.GT.0.9999) GO TO 1720
       N = N+1
       GO TO 1700
C
 1720  Nmaxti=N
       FVtin(Nmaxti)=1.0
       Pmaxti=P
C
       J=0
       DO 1800 N=1,Nmaxti
       J = J+1
       IF(J.LT.10) GO TO 1750
       J=0
       PAUSE 'PAUSE DURING PRINT'
1750   P= Pmini + N*DPi - DPhalfi
1800   WRITE(0,9) N,P,FVtin(N)

C  INTERPOLATE THE INHALATION CURVE TO SAME NMAX AS THE EXHALATION CURVE
       CALL INTERP(FVtin,Nmaxti,DPi,Nmaxte,DPe)

C PRINT THE DATA TO AN OUTPUT FILE

 15    FORMAT (1X,I4,3F9.2)
 16    FORMAT(1X,F7.3,1X,F7.3)

       Pminti=Pmini+DPhalfe
       Pmaxti=Pminti+(Nmaxte-1)*DPe
```

```
      WRITE(LUOUT,15) Nmaxte,Pminti,Pmaxti,DPhys

      DO 1900 N=1,Nmaxte
      P= Pmini + N*DPe - DPhalfe
1900  WRITE(LUOUT,16) P,FVtin(N)

      DO 2000 N=1,Nmaxte
      P= Pmine+ N*DPe - DPhalfe
 2000 WRITE(LUOUT,16) P,FVtex(N)
C
      END
C_______________________________________________________________

C  THE ERROR FUNCTION
      FUNCTION ERF(X)
      IF(X.LT.0.)THEN
        ERF=-GAMMP(.5,X**2)
      ELSE
        ERF=GAMMP(.5,X**2)
      ENDIF
      RETURN
      END
C_______________________________________________________________
C
C THE GAMMA FUNCTION
      FUNCTION GAMMP(A,X)
      IF(X.LT.0..OR.A.LE.0.)PAUSE
      IF(X.LT.A+1.)THEN
      CALL  GSER(GAMSER,A,X,GLN)
        GAMMP=GAMSER
      ELSE
        CALL GCF(GAMMCF,A,X,GLN)
        GAMMP=1.-GAMMCF
      ENDIF
      RETURN
      END
C_______________________________________________________________
C
      SUBROUTINE GSER(GAMSER,A,X,GLN)
      PARAMETER (ITMAX=100, EPS=3.E-7)
      GLN = GAMMLN(A)
      IF(X.LE.0.)THEN
        IF(X.LT.0.)PAUSE
        GAMSER=0.
        RETURN
      ENDIF
      AP=A
      SUM=1./A
      DEL=SUM
      DO 11 N=1,ITMAX
        AP=AP+1.
        DEL=DEL*X/AP
        SUM=SUM+DEL
        IF(ABS(DEL).LT.ABS(SUM)*EPS)GO TO 1
11    CONTINUE
      PAUSE 'A too large, ITMAX too small'
1     GAMSER = SUM*EXP(-X+A*LOG(X)-GLN)
      RETURN
      END
C_______________________________________________________________
C
      SUBROUTINE GCF(GAMMCF,A,X,GLN)
      PARAMETER (ITMAX=100,EPS=3.E-7)
      GLN=GAMMLN(A)
      GOLD=0.
```

Program TISSFCM.FOR

```
      A0=1.
      A1=X
      B0=0.
      B1=1.
      FAC=1.
      DO 11 N=1,ITMAX
        AN=FLOAT(N)
        ANA=AN-A
        A0=(A1+A0*ANA)*FAC
        B0=(B1+B0*ANA)*FAC
        ANF=AN*FAC
        A1=X*A0+ANF*A1
        B1=X*B0+ANF*B1
        IF(A1.NE.0.)THEN
          FAC=1./A1
          G=B1*FAC
          IF(ABS((G-GOLD)/G).LT.EPS)GO TO 1
          GOLD=G
        ENDIF
11    CONTINUE
      PAUSE 'A too large, ITMAX too small'
1     GAMMCF=EXP(-X+A*ALOG(X)-GLN)*G
      RETURN
      END
C______________________________________________________________
C
      FUNCTION GAMMLN(XX)
      REAL*8 COF(6),STP,HALF,ONE,FPF,X,TMP,SER
      DATA COF,STP/76.18009173D0,-86.50532033D0,24.01409822D0,
     *    -1.231739516D0,.120858003D-2,-.536382D-5,2.50662827465D0/
      DATA HALF,ONE,FPF/0.5D0,1.0D0,5.5D0/
      X=XX-ONE
      TMP=X+FPF
      TMP=(X+HALF)*LOG(TMP)-TMP
      SER=ONE
      DO 11 J=1,6
        X=X+ONE
        SER=SER+COF(J)/X
11    CONTINUE
      GAMMLN=TMP+LOG(STP*SER)
      RETURN
      END
C______________________________________________________________
C
      SUBROUTINE IODECLS(INP, OUT, PLT)
      INTEGER INP, OUT, PLT
      CHARACTER*1 BEEP
      CHARACTER*80 ITITLE
      CHARACTER*64 NAINP,NAOUT,NAPLT
      LOGICAL IEXIST
      COMMON /DEVS/ LUTRM, LUIN, LUOUT, LUPLT
      COMMON /CURSOR/ BEEP
      COMMON /LABELS/ ITITLE, NAINP, NAOUT, NAPLT
      QUOTE = '"'
      LUTRM = 0
      LUIN = 1
      LUOUT = 2
      LUPLT = 3
C
C    TERMINAL:  LOGICAL UNIT 0
C       INPUT:  LOGICAL UNIT 1  (DATA FILE)
C      OUTPUT:  LOGICAL UNIT 2
C   PLOT FILE:  LOGICAL UNIT 3
C
      IEXIST =.FALSE.
```

Program TISSFCM.FOR

```
      IF (INP .NE. 0) THEN
         WRITE(LUTRM,'(//,5X,''ENTER INPUT FILE NAME.'')')
    5    READ(LUTRM,'(A64)') NAINP
         INQUIRE(FILE=NAINP,EXIST=IEXIST)
         IF (IEXIST) THEN
            OPEN(LUIN,FILE=NAINP,STATUS='OLD')
            WRITE(LUTRM,'(//,5X,''INPUT FILE NAME: '',A64)') NAINP
          ELSE
            WRITE(LUTRM,'(5X,''*** FILE NAME:  '',A64,/,5X,A1,
     *      ''NOT FOUND'')')  NAINP,BEEP
            WRITE(LUTRM,'(/,5X,''ENTER A NEW FILE NAME.'')')
            GOTO 5
          END IF
      END IF
      IEXIST = .FALSE.
      IF (OUT .NE. 0) THEN
          WRITE(LUTRM,'(/,5X,''ENTER OUTPUT FILE NAME.'')')
   10     READ(LUTRM,'(A64)') NAOUT
          INQUIRE(FILE=NAOUT,EXIST=IEXIST)
        IF(IEXIST) THEN
          WRITE(LUTRM,'(5X,''*** FILE NAME:  '',A64,/,5X,A1,
     *    ''ALREADY EXISTS.  DO YOU WISH TO OVERWRITE IT? (Y/N)'')')
     *    NAOUT,BEEP
          READ(LUTRM,'(A1)') ANS
          IF(ANS .EQ. 'Y' .OR. ANS .EQ. 'y') THEN
             OPEN(LUOUT,FILE=NAOUT,IOSTAT=IERR,STATUS='OLD')
             REWIND LUOUT
          ELSE
             WRITE(LUTRM,'(/,5X,''ENTER A NEW FILE NAME'')')
             GOTO 10
          END IF
        ELSE
          OPEN(LUOUT,FILE=NAOUT,IOSTAT=IERR,STATUS='NEW')
        END IF
        WRITE(LUTRM,'(//,5X,''OUTPUT FILE NAME: '',A64)') NAOUT
        WRITE(LUTRM,'(/,5X,''DO YOU WISH TO WRITE A TITLE LINE ON THE'',
     *  '' OUTPUT FILE? (Y/N)'')')
        READ(LUTRM,'(A1)') ANS
        IF(ANS .EQ. 'Y' .OR. ANS .EQ. 'y') THEN
           WRITE(LUTRM,'(/,5X,''ENTER A TITLE LINE.'')')
           READ(LUTRM,'(A80)') ITITLE
           WRITE(LUOUT,'(A1,A80,A1)') QUOTE,ITITLE,QUOTE
         END IF
      END IF
CFF
      IEXIST = .FALSE.
      IF (PLT .NE. 0) THEN
          WRITE(LUTRM,'(/,5X,''ENTER OUTPUT PLOT FILE NAME.'')')
   15     READ(LUTRM,'(A64)') NAPLT
          INQUIRE(FILE=NAPLT,EXIST=IEXIST)
        IF(IEXIST) THEN
          WRITE(LUTRM,'(5X,''*** FILE NAME:  '',A64,/,5X,A1,
     *    ''ALREADY EXISTS.  DO YOU WISH TO OVERWRITE IT? (Y/N)'')')
     *    NAPLT,BEEP
          READ(LUTRM,'(A1)') ANS
          IF(ANS .EQ. 'Y' .OR. ANS .EQ. 'y') THEN
             OPEN(LUPLT,FILE=NAPL‾,IOSTAT=IERR,STATUS='OLD')
             REWIND LUPLT
          ELSE
             WRITE(LUTRM,'(/,5X,''ENTER A NEW FILE NAME'')')
             GOTO 15
          END IF
        ELSE
          OPEN(LUPLT,FILE=NAPLT,IOSTAT=IERR,STATUS='NEW')
        END IF
```

```
      WRITE(LUTRM,'(//,5X,''PLOT FILE NAME: '',A64)') NAPLT
    END IF
    RETURN
    END
C------------------------------------------------------------------
      SUBROUTINE INTERP(Y,IMAX1,DX1,IMAX2,DX2)
    DIMENSIONY(211), X1(211), X2(211), Y2(211)
    DO 10 I1 = 1,IMAX1
       X1(I1) = (I1 -1)*DX1
 10   CONTINUE
    DO 20 I2 = 1,IMAX2
       X2(I2) = (I2-1)*DX2
 20   CONTINUE
    DO 30 I2 = 1,IMAX2
       I1 = X2(I2)/DX1 + DX1/2.0
       I1 = I1 +1
       IF(I1 .EQ. 1) I1 = 2
       IF(I1 .EQ. IMAX1) I1 = IMAX1 - 1
       U = (X2(I2) - X1(I1))/DX1
       D1 = Y(I1+1) - Y(I1-1)
       D2 = Y(I1+1) - 2.0*Y(I1) + Y(I1-1)
       Y2(I2) = Y(I1) + 0.5*D1*U + 0.5*D2*U**2
 30   CONTINUE
    DO 40 I2 = 1,IMAX2
       Y(I2) = Y2(I2)
 40   CONTINUE
    RETURN
    END
```

Appendix C - 2
Tissue Model - Part 2

PROGRAM REGPVCUR

```
C  THIS PROGRAM DETERMINES THE REGIONAL P-V CURVES FOR THE REGIONS LOCATED
C       AT DIFFERENT VERTICAL POSITIONS
      DIMENSION V(4,200),VO(10),ZQ(10),DV(10),VGAS(10),Ppar(2,201),
     *     POSN(20), PCG(4,200), RFC(4,200), Pst(2,4,200),DPGRAVO(10)
      COMMON A,B,NMAX,NMAX1,CHECKV,DNSTY,IMAX, IMAX1, PSTMIN,
     *  PSTMAX, EBAR, VE(101), Z(101), ZI(101), ZO(101), SMASS(101),
     *     P(101),PO(101),DENS(101),E(201),CHECKP,DP,XMASS(101),
     *  VETIS(201),VMV(101),VLLUNG(301),VEO(51),VETV(51),
     *  CVG3,FI(51),PMIN,TVE,XAZMAX,ZMAX,SA(101),TSMASS(101),
     *  BLMASS(101),H(101),PCAP(101),HRTPOS,HO,PSI,
     *     PART,K,ZMAX2,FC(2,201)
      INTEGER R
      INTEGER INP, OUT, PLT
      INTEGER CRT, LUOUT, LUIN, LUPLT
      CHARACTER*1 BEEP, ANS, QUOTE
      CHARACTER*80 ITITLE
      CHARACTER*64 NAINP,NAOUT,NAPLT
      LOGICAL IEXIST
      COMMON /DEVS/LUTRM, LUIN, LUOUT, LUPLT
      COMMON /CURSOR/ BEEP
      COMMON /LABELS/ITITLE,NAINP,NAOUT,NAPLT

    1 FORMAT(I2,2F8.2)
    2 FORMAT(1X,'FC(Jstatus,I) ',10F8.4)
    3 FORMAT(6F20.3)
    4 FORMAT(10X,I3,4F18.3)
    5 FORMAT(5X,I4,2F10.3)
    6 FORMAT(1X,'PMIN PSTMAX DPHYS IM',3F8.3,I3)
    7 FORMAT('IPOS = ',I3)
    8 FORMAT(' VOLUME INCREMENT = ',I3)
1101  FORMAT(3X,'K = ',I3,3X,' LUNG VOL = ',F9.2)
C
C  ISTAT-1 IS A STATIC N2 WASHIN-WASHOUT
C  ISTAT = 2 IS A STUDY OF THE STATIC VOLUME CHANGE OF FOUR
C  REGIONS OF EQUAL MASS
C
      ISTAT = 2
C     ISTAT = 1
C
C  VETIS(I) IS READ IN AS THE GAS FRACTION OF CAPACITY AND MUST BE
C  CONVERTED TO THE VOLUME EXPANSION OF THE TISSUE,BETIS(I).
C
      CRT = 0

C  READ THE DATA FILE WITH THE PARENCHYMAL INHALATION AND EXHALATION
C  P-FV CURVES THAT HAVE BEEN GENERATED IN PROGRAM TISSFCM.  THESE
C  CURVES INCLUDE THE AFFECTS OF TERMINAL BRONCHI CLOSING AND OPENING
C  ON THE PARENCHYMAL ELASTIC CHARACTERISTICS.
C

      CALL IODECLS(1, 0, 0)
  15  FORMAT(1X,I4,3F9.2)
  16  FORMAT(1X,F7.3,1X,F7.3)

      READ (LUIN,15) IM,Pmin,Pstmax,DPhys
C      WRITE(CRT,15) IM,Pmin,Pstmax,DPhys
C NOTE: Pstmin IS DEFINED IN SUBROUTINE VARYMNV

C      PAUSE 'PAUSE DURING THE READING OF THE DATA'
C
```

Program REGPVCUR.FOR

```
C       JSTATUS=1 INDICATES THE INHALATION P-FV CURVE
C       JSTATUS=2 INDICATES THE EXHALATION P-FV CURVE
C
      Jstatus=1
      DO 18 N=1,IM
      READ(LUIN,16) Ppar(Jstatus,N),FC(Jstatus,N) !Inhalation Ppar-FV curve
 18   CONTINUE
C
      Jstatus=2
      DO 19 N=1,IM
      READ(LUIN,16) Ppar(Jstatus,N),FC(Jstatus,N) !Exhalation Ppar-FV curve
 19   CONTINUE
C
C     DO 40 Jstatus=1,2
C     NC=0
C     DO 40 N=1,IM
C     NC=NC+1
C     IF(NC.LT.22) GOTO 40
C     NC=0
C     PAUSE'PAUSE TO LOOK AT Ppar AND FC FROM TISPFV'
C   40     WRITE(CRT,16) Ppar(Jstatus,N),FC(Jstatus,N)
C
C   ANATOMICAL DEAD SPACE [ML]
C
      ADSP = 160.00
      PDS2 = ADSP/2.0
      PDS4 = ADSP/4.0
      IMAX = 201
      IMAX1 = IMAX - 1
C
C  EXPANSION INFORMATION
C
      ALPHA = 0.20
      BETA = 0.30
      GAMMA = 0.50
C
C  LUNG VOLUME, MASS AND GEOMETRY INFORMATION
C  VMAX AND AMV REFER TO THE VOLUME OF TISSUE AND GAS.  RV, FRC
C  AND TLC REFER TO MEASURED GAS VOLUMES.
C
C   RESIDUAL VOLUME [ML]
C
        RV = 2000.0     !NRA
C       RV = 1890.00    !JCC
C       RV = 1000.00    !TEST CASE
C
C   FUNCTIONAL RESIDUAL CAPACITY [ML]
C
        FRC = 3810.00           !NRA
C       FRC = 3400.00           !JCC
C       FRC = 2500.00           !TEST CASE
C
C   TOTAL LUNG CAPACITY [ML]
C
        TLC = 7600.00           !NRA
C       TLC = 6400.00           !JCC
C       TLC = 4000.00           !TEST CASE
C
C  CORRECT VOLUMES FOR ANATOMICAL DEAD SPACE
C
       RV = RV - ADSP
       FRC = FRC - ADSP
       TLC = TLC - ADSP
C
C      GRAMS = MASS OF LUNG = OVERALL DENSITY * TOTAL LUNG CAPACITY;
```

Program REGPVCUR.FOR

```
C         THE DENSITY AT TLC IS ASSUMED TO BE 0.13333gm/cc
C
      GRAMS = TLC*0.13333
C
C     GET VOLUMES AND WEIGHT FOR ONE LUNG
C
      GRAMS = GRAMS/2.0
      RV = RV/2.0
      FRC = FRC/2.0
      TLC = TLC/2.0
      VC=TLC-RV                 !Vital Capacity for a single lung
      VMAX = TLC + GRAMS
      Jstatus=1                 !Inhalation status
      FCMIN = FC(Jstatus,1)

      AMV = FCMIN*TLC + GRAMS !Amount Minimum Volume = smallest air
c                              vol + tissue vol outside chest.
C
C  DENSITY = MASS/VOLUME
C
      DNSTY = GRAMS/AMV
C      WRITE(0,3) DNSTY,AMV,GRAMS
C      PAUSE 'PAUSE AFTER DNSTY,AMV AND GRAMS ARE PUT ON THE SCREEN'

      TC = AMV/1250.
      A0 = 7.00*TC**ALPHA
      B0 = 5.25*TC**BETA
      C0 = 10.50*TC**GAMMA
      THETA = 3.1416*A0*B0/C0**2
C
C  CALCULATE THE TISSUE FC-P RELATION FOR THE SUBJECT.  THE TISSUE
C  FCMIN IS BASED ON THE RATIO RV/T1C
C
      IPOS = 2
C      WRITE(CRT,7) IPOS
C      PAUSE'PAUSE BEFORE YOU ENTER  VRYMNV FOR THE FIRST TIME'
      CALL VRYMNV(FCMIN,TLC,GRAMS,IM,AMV,Jstatus)
      IPOS = 3
C      WRITE(CRT,7) IPOS
C
C  CALCULATION OF ZMAX FOR THE GIVEN SHAPE AND GAS AND TISSUE
C  VOLUME.
C
 65   FORMAT (2X,' ZMAX0 = ',F7.2)
      G1 = 1.0
      G2 = -C0*3.0
      G3 = 0.0
      G4 = 3.0*AMV/THETA
      Z1 = 20.0
      CALL SOLVEZ(G1,G2,G3,G4,Z1,ZMAX0)
      WRITE(CRT,65) ZMAX0
C      PAUSE'PAUSE AFTER YOU WRITE ZMAX0'
C
C   INFORMATION THE BREATH BEING ANALYZED
C
      WRITE(CRT,3) RV, FRC, TLC
C
      VSTART = FCmin*TLC       ! Vstart=lung volume inhalation begins
      VSTOP =  FCmin*TLC       ! Vstop=lung volume exhalation ends
      TIDVOL=TLC*(1.0-FCMIN)   ! TIDal VOLume for this breath
C
C  HEART AND BLOOD VOLUME INFORMATION
C
      PART = 10.00
      HRTPOS = 15.00
```

Program REGPVCUR.FOR

```
      SAPGT = 30000.00
      NO = 0.00000428
      PSI = 0.000000219
C
C  NUMBER OF LUNG SECTIONS
C
      NMAX = 51
      NMAX1 = NMAX - 1
      DZ = ZMAX0/NMAX1
C
C     CONDITIONS WHEN THE LUNG IS AT MINIMUM VOLUME
C
      ZI(1) = 0.0
      Z(1) = 0.0
      XMASS(1) = 0.0
      SMASS(1) = 0.0
      VMV(1) = 0.0
      VE(1) = 0.
      JCOUNT=0
      DO 20 N = 2,NMAX
       JCOUNT=JCOUNT+1
        VE(N) = 1.0
        ZI(N) = (N-1)*DZ
        VMV(N) = THETA * (CO * (ZI(N)**2 - (ZI(N-1))**2) -
     *        (ZI(N)**3 - ZI(N-1)**3)/3.0)
        TSMASS(N) = DNSTY * VMV(N)
        SA(N) = SAPGT*TSMASS(N)
       XMASS(N) = TSMASS(N) + XMASS(N-1)
C      WRITE(0,21)N,VMV(N),TSMASS(N),SA(N),XMASS(N)
       IF(JCOUNT.LT.20) GOTO 20
       JCOUNT=0
C      PAUSE' PAUSE TO LOOK AT N,VMV(N),TSMASS(N),SA(N),XMASS(N)'
   20 CONTINUE
   21 FORMAT(2X,I5,4(2X,F8.3))
C     WRITE(CRT,3) AMV, XMASS(NMAX)
C     PAUSE'PAUSE AFTER VMV PRINT OUT'

       RWEIGHT = GRAMS/4.0       !Tissue weight of a region in gms
       RTLC = TLC/4.0            !Regional gas volume in cc
      DO 25 L = 1, 4
       V(L,1) = AMV/4.0
   25 CONTINUE

C    DETERMINE POSN(J), ODD VALUES OF J DEFINE THE POSITION OF THE CG
C    ON THE MATERIAL COORDINATE SYSTEM, EVEN VALUES DEFINE THE LOWER
C    SURFACE OF EACH REGION ON THE MATERIAL COORDINATE SYSTEM.
C
      FRSIZE=.125
      K=1
      FRMASS=0
      ICOUNT=0

      DO 30 N = 2,NMAX
       FRMASSO=FRMASS
       FRMASS = XMASS(N)/XMASS(NMAX)
       IF(FRMASS.LT.K*FRSIZE) GOTO 26
       POSN(K)=N-1+(K*FRSIZE-FRMASSO)/(FRMASS-FRMASSO)
       K=K+1
   26  CONTINUE
C      WRITE(CRT,5) N-1, XMASS(N), FRMASS
       ICOUNT=ICOUNT+1
     . IF(ICOUNT.NE.17) GOTO 30
       ICOUNT=0
C      PAUSE'PAUSE TO LOOK AT THE MASS DISTRIBUTION'
   30 CONTINUE
```

Program REGPVCUR.FOR

```
C      PAUSE'PAUSE TO LOOK AT THE MASS DISTRIBUTION'

       POSN(8)=  NMAX + .000001

       K1= POSN(1)
       K2= POSN(2)
       K3= POSN(3)
       K4= POSN(4)
       K5= POSN(5)
       K6= POSN(6)
       K7= POSN(7)
       K8= POSN(8)
C
C  CALCULATE THE DPGRAV); THE PRESSURE DIFFERENCE BETWEEN THE REGIONAL
C                CG's AT THE MINIMUM VOLUME
C
       DPGRAVO(1)= (POSN(3)-POSN(1))*DZ*DNSTY
       DPGRAVO(2)= (POSN(5)-POSN(3))*DZ*DNSTY
       DPGRAVO(3)= (POSN(7)-POSN(5))*DZ*DNSTY
C
C      DO 31 K=1,8
C 31   WRITE(CRT,5) K,POSN(K),DPGRAVO(K)
C 32   FORMAT(5X,8I6)
C      PAUSE'PAUSE AFTER THE PRINTING OF THE POSN(K),DPGRAVO'
C
C  ITERATION INFORMATION
C
      CHECKV = 2.0
      CHECKP = 0.01
      DPI = 0.5
      P2 = 4.00
C
C   CALCULATE THE NUMBER OF VOLUME INCREMENTS (DVOL) FOR THIS BREATH.
C   THE LAST ONE IS K End EXhalation (KEEX). THE SIZE OF DVOL IS
C   ESTIMATED THEN ADJUSTED SLIGHTLY SO THAT IT FITS EVENLY INTO THE
C   TIDAL VOLUME (TIDVOL).
C
       DVOL=40.00
       KEINSM1=TIDVOL/DVOL          ! KEINS-1
       DVOL=TIDVOL/KEINSM1          ! Recalculate DVOL
       KEINS=KEINSM1+1              ! Calculate K End INspiration
       KEINSP1=KEINS+1              ! KEINS+1
       KEEX=KEINS+((VSTART+TIDVOL)-VSTOP)/DVOL!Calculate KEEX
       KEEXM1=KEEX-1
       KEEXM2=KEEX-2
       KEEXP1=KEEX+1
 35   FORMAT(4X,5F10.2)
C      WRITE(0,35) VLLUNG(1),VSTART,GRAMS,TIDVOL,DVOL
C      PAUSE'PAUSE AFTER PRINTING VLLUNG(1),VSTART,GRAMS,TIDVOL,DVOL'

      VLLUNG(1) = VSTART + GRAMS

      DO 1020 K = 2,KEINS
 1020 VLLUNG(K) = VLLUNG(K-1) + DVOL
      DO 1025 K=KEINSP1,KEEX
 1025 VLLUNG(K) =VLLUNG(K-1) -DVOL

      DO 1050 L=1,4
 1050 VO(L)=V(L,1)

C
C-------------------MAIN TIME DO LOOP--------------------------------
C
       K=0
 5000  K=K+1
```

```
      IF(K.GT.KEEX) GOTO 500

      WRITE(CRT,1101) K,VLLUNG(K)
C     PAUSE'PAUSE BEFORE THE Kth LUNG VOLUME CALC STARTS'

      IF(K.NE.KEINSP1) GOTO 100
C
C  CONVERT THE EXHALATION P-FV CURVE TO A P-VE (Volume Expansion) CURVE
C
      GOTO 2008
   99 CONTINUE
      Jstatus=2
      Pmin=Pmin-DPhys
      Pstmax=Pstmax-DPhys
      CALL VRYMNV(FCNIN,TLC,GRAMS,IN,AMV,Jstatus)
      P2=P2-DPhys

  100 CONTINUE
        VOL = VLLUNG(K)
        VOLEXP = VOL/AMV
        ZE = VOLEXP**GAMMA
        ZMAX = ZMAX0*ZE
        ZMAX2 = ZMAX + 2.0
        TVE = THETA*VOLEXP**(ALPHA + BETA - 2.0*GAMMA)
        CVG = C0*VOLEXP**GAMMA
        CVG3 = CVG*3
        XAZMAX = TVE*(2.0*CVG*ZMAX - ZMAX**2)
C
C FIRST APPROX FOR DENS(N) AND Z(N) - USE THE VALUES FOR THE LUNG
C IN THE UNWEIGHTED STATE - AS WEST STARTED HIS OUT.
C
        IF(K .EQ. 1) THEN
         DO 110 N = 2,NMAX
            DENS(N) = DNSTY/VOLEXP
            Z(N) = ZE*ZI(N)
  110   CONTINUE
        END IF
      DO 120 N = 2,NMAX
           VEO(N) = VE(N)
           ZO(N) = Z(N)
  120 CONTINUE
        VCALO = VCALC
C
C FOR THE GIVEN P2,DENS,Z AND GEOMETRY (R**2 = A*(1 + EBAR)*2/PI*Z,
C FIND THE VOLUME-VCALC
C
C      PAUSE'PAUSE BEFORE THE FIRST CALL DPDG'
      CALL DPDG(P2, VCALC)
C     WRITE(0,3) P2, VOL, VCALC
        F2 = VOL - VCALC
        IF(F2 .LT. 0.0) THEN
            P3 = P2 - DPI
          ELSE
            P3 = P2 + DPI
        END IF
  130 F1 = F2
        P1 = P2
      P2 = P3
C     PAUSE'PAUSE BEFORE THE SECOND CALL DPDG'
      CALL DPDG(P2,VCALC)
        F2 = VOL - VCALC
      AF2 = ABS(F2)
      WRITE(CRT,51) P1,P2,F1,F2,VOL,VCALC
   51 FORMAT(2X,6F9.2)
      IF(AF2 .LT. CHECKV) GO TO 140
```

Program REGPVCUR.FOR

```
        P3 = (P1*F2 - P2*F1)/(F2-F1)
      GO TO 130
 140  CONTINUE

      ICOUNT=0
 101  FORMAT(2X,I3,3F10.2)
      DO 150 N=1,NMAX
C     WRITE(0,101) N,P(N),VE(N),VMV(N)
      ICOUNT=ICOUNT + 1
      IF(ICOUNT.NE.17) GO TO 150
C     PAUSE'PAUSE TO CHECK N,P(N),VE(N),VMV(N) DATA'
      ICOUNT=0
 150  CONTINUE
C
C CALCULATE THE VOLUME AND THE STATIC RECOIL PRESSURE AT THE CG OF EACH
C          OF THE FOUR REGIONS IN THE DYNAMIC MODEL
C
C
C                        REGION 1
      V(1,K) = 0.0
      DO 220 N = 2,K2
         V(1,K) = V(1,K) + VE(N)*VMV(N)
 220  CONTINUE
      V(1,K) = V(1,K) +(POSN(2)-K2)*VE(K2+1)*VMV(K2+1)
      PCG(1,K)= P(K1) + (POSN(1)-K1)*(P(K1+1)-P(K1))
C
C                        REGION 2

      V(2,K) = (K2+1-POSN(2))*VE(K2+1)*VMV(K2+1)
      DO 230 N = K2+2,K4
         V(2,K) = V(2,K) + VE(N)*VMV(N)
 230    CONTINUE
      V(2,K) = V(2,K) + (POSN(4)-K4)*VE(K4+1)*VMV(K4+1)
      PCG(2,K)= P(K3) + (POSN(3)-K3)*(P(K3+1)-P(K3))
C
C                        REGION 3
      V(3,K) = (K4+1-POSN(4))*VE(K4+1)*VMV(K4+1)
      DO 240 N = K4+2,K6
         V(3,K) = V(3,K) + VE(N)*VMV(N)
 240  CONTINUE
      V(3,K)=V(3,K)+(POSN(6)-K6)*VE(K6+1)*VMV(K6+1)
      PCG(3,K)= P(K5) + (POSN(5)-K5)*(P(K5)-P(K5+1))
C
C                        REGION 4
      V(4,K) = (K6+1-POSN(6))*VE(K6+1)*VMV(K6+1)
      DO 250 N=K6+2,NMAX
         V(4,K) = V(4,K) + VE(N)*VMV(N)
 250  CONTINUE
      PCG(4,K)= P(K7) + (POSN(7)-K7)*(P(K7)-P(K7+1))

      VOLCHG = VCALC - VCALO
C     WRITE(0,4)
 255  FORMAT(2X,I3,5F10.3)
        DO 260 L = 1,4
         VGAS(L) = V(L,K) - RWEIGHT
         DV(L) = V(L,K) -VO(L)
         ZQ(L) = DV(L)/VOLCHG
         WRITE(CRT,255) L,V(L,K),PCG(L,K),VGAS(L), DV(L), ZQ(L)
 260  CONTINUE

      GASVOL = 2.0*(VLLUNG(K) - GRAMS) + ADSP
        DO 265 L = 1,4
         VREG = 2.0*VGAS(1) + PDS4
         VO(L)=V(L,K)
 265    CONTINUE
```

Program REGPVCUR.FOR

```
      GOTO 5000
C-----------------------END TO TIME LOOP-----------------------------
C
  500 CONTINUE
C
C  PRINT PROGRAM OUTPUT TO A NAMED DATA FILE FOR BY THE DYNamic
C  REGional  flow (Q) PROGRAM (PRG DYNREGQ).  THE DATA WILL REPRESENT
C  THE REGIONAL LUNG CHARACTERISTICS FOR A SPECIFIC SUBJECT.
C
C             NAME THE OUTPUT FILE
C
      CALL IODECLS(0,1,0)
C  RECONSTRUCT RV,FRC,TLC,AMV AND GRAMS FOR THIS SUBJECT
C
      RV=2*RV+ADSP
      FRC=2*FRC+ADSP
      TLC=2*TLC+ADSP
      GRAMS=2*GRAMS
      AMV=2*AMV
C
C  WRITE THE RV,FRCTLC,AND ADSP FOR THIS SUBJECT TO THE NAMED FILE
      WRITE(LUOUT,2005) RV,FRC,TLC,ADSP
C
C  WRITE THE EXPANSION COEFFICIENTS, LUNG VOLUME(GAS AND TISSUE) AT
C  MINIMUM VOLUME AND THE TISSUE WEIGHT OF THE LUNGS.
C
      WRITE(LUOUT,2005)ALPHA,BETA,AMV,GRAMS
C
C  WRITE OUT THE PRESSURE DIFFERENCES BETWEEN THE CG's OF THE FOUR
C  REGIONS WHEN THEY ARE AT MINIMUM VOLUME.
C
      WRITE(LUOUT,2005) (DPGRAVO(I),I=1,3)
C
C WRITE OUT THE EXHALATION P-FV CURVE; FIRST DETERMINE THE REGIONAL Pst-
C FV RELATION AS A FUNCTION OF A CONSTANT DFC.THIS RELATIONSHIP WILL
C BE INPUT INTO PRG DYNREGQ.
C
 2008  CONTINUE
       JMAX=50
       JMAXM1=JMAX-1
       FCMAX=1.0
       DFC=(FCMAX-FCMIN)/JMAXM1
C
 2001 FORMAT(5X,' REGION =',I2)
 2002 FORMAT(3X,2I5,4F10.2)
 2003 FORMAT(I3,3F10.4)
 2004 FORMAT(10F8.4)
 2005 FORMAT(5F10.4)
 2006 FORMAT(3X,I5,4F10.3)
C
      INCREMENT=1
      IF(Jstatus.EQ.2) INCREMENT=-1
      KJBEG=1
      IF(Jstatus.EQ.2) KJBEG=KEEX
      KJEND=KEINS
      IF(Jstatus.EQ.2) KJEND=KEINS
C
C  CONVERT THE REGIONAL GAS VOLUMES INTO REGIONAL GAS FRACTIONS (RFC)
C
      DO 2050 R=1,4
      WRITE(CRT,2001) R
      ICOUNT=0
      DO 2010 KJ=KJBEG,KJEND,INCREMENT
      ICOUNT=ICOUNT+1
      RFC(R,KJ)=(V(R,KJ)-RWEIGHT)/RTLC
```

Program REGPVCUR.FOR

```
      WRITE(CRT,2006) KJ, V(R,KJ), RFC(R,KJ), PCG(R,KJ), RTLC
      IF(ICOUNT.LT.20) GOTO 2010
      PAUSE'PAUSE TO LOOK AT  KJ, V(R,KJ), RFC(R,KJ), PCG(R,KJ), RTLC'
      ICOUNT=0
 2010 CONTINUE
      PAUSE'PAUSE TO LOOK AT  KJ, V(R,KJ), RFC(R,KJ), PCG(R,KJ), RTLC'
C
C  DETERMINE THE Pst AS A FUNCTION OF RFC IN INCREMENTS OF DFC.J IS
C  THE VARIABLE FOR THE NEW Pst-FC DATA FOR PRG DYNREGQ.  KJ IS THE
C  VARIABLE FOR THE REGIONAL Pcg-RFC DATA THAT REPRESENTS THE
c  VOLUME INCREMENTS DURING THE INHALATION AND EXHALATION. THE
C  CURVE INTERPOLATION STARTS AT FCMIN (J=1)AND INCREASES IN DFC
C  INCREMENTS, SO THE INITIAL VALUE FOR KJ, KBEG,DEPENDS ON
C  WHETHER IT IS FOR INHALATION OR EXHALATION.
C
       J=1
       KJ=KJBEG
       ICOUNT=0
 2020  ICOUNT=ICOUNT+1
       FCJ=FCMIN+(J-1)*DFC
       IF(J.EQ.1) GOTO 2035
 2030  IF(RFC(R,KJ).GT.FCJ) GOTO 2040
       IF(KJ.EQ.KJEND) GOTO 2040
       KJ=KJ+INCREMENT
       GOTO 2030
 2035  CONTINUE
C
C  CALCULATE THE PST FOR THE FIRST POINT ON THE PST CURVE (J=1)
C
       KJ1=KJ
       KJ2=KJ+INCREMENT
 2037  IF(RFC(R,KJ2).GT.RFC(R,KJ1)) GOTO 2038
       KJ1=KJ1+INCREMENT
       KJ2=KJ2+INCREMENT
       GOTO 2037
 2038  U=(FCJ-RFC(R,KJ1))/(RFC(R,KJ2)-RFC(R,KJ1))
       PST(Jstatus,R,J)=Pcg(R,KJ1)+U*(Pcg(R,KJ2)-PCG(R,KJ1))
      WRITE(CRT,2002) J,KJ,FCJ,Pst(Jstatus,R,J),PCG(R,KJ),RFC(R,KJ)
       KJ=KJ2
       J=J+1
       GOTO 2020
 2040  U=(FCJ-RFC(R,KJ-1))/(RFC(R,KJ)-RFC(R,KJ-1))
       PST(Jstatus,R,J)=Pcg(R,KJ-1)+U*(Pcg(R,KJ)-Pcg(R,KJ-1))
      WRITE(CRT,2002) J,KJ,FCJ,Pst(Jstatus,R,J),PCG(R,KJ),RFC(R,KJ)
       IF(ICOUNT.LT.22) GOTO 2045
      PAUSE'PAUSE TO SEE J,KJ,FCJ,Pst(Jstatus,R,J),PCG(R,KJ),RFC(R,KJ)'
       ICOUNT=0
 2045  IF(J.EQ.JMAX)GOTO 2048
       J=J+1
       GOTO 2020
 2048  CONTINUE
      PAUSE'PAUSE TO SEE J,KJ,FCJ,Pst(Jstatus,R,J),PCG(R,KJ),RFC(R,KJ)'
 2050  CONTINUE
       IF(JSTATUS.EQ.1) GOTO 99
C
C  WRITE PST DATA TO A NAMED FILE FOR USE IN PROGRAM DYNREGQ
C
      WRITE(LUOUT,2003) JMAX,FCMIN,FCMAX
      DO 2100 R=1,4
      DO 2100 Jstatus=1,2
 2100 WRITE(LUOUT,2004)(Pst(Jstatus,R,J), J=1,JMAX)
     . WRITE(LUOUT,2003) IM,Pmin,Pstmax,DPhys
      WRITE(LUOUT,2004)(FC(1,N),N=1,IM)    !Original data from TISPV
      WRITE(LUOUT,2004)(FC(2,N),N=1,IM)    !Original data from TISPV
      STOP
```

```
      END
C-------------------------------------------------------------------
      SUBROUTINE DPDG(P2,VCALC)
C Subroutine calculates the tissue pressures at the top of each section
c    for the given density using the hydrostatic pressure equation
        COMMON A,B,NMAX,NMAX1,CHECKV,DNSTY,IMAX, IMAX1, PSTMIN,
     *  PSTMAX, EBAR, VE(101), Z(101), ZI(101), ZO(101), SMASS(101),
     *     P(101),PO(101),DENS(101),E(201),CHECKP,DP,XMASS(101),
     *  VETIS(201),VMV(101),VLLUNG(301),VEO(51),VETV(51),
     *  CVG3,FI(51),PMIN,TVE,XAZMAX,ZMAX,SA(101),TSMASS(101),
     *  BLMASS(101),H(101),PCAP(101),HRTPOS,HO,PSI,
     *     PART,K,ZMAX2,FC(2,201)
 10   FORMAT(3X,3F12.3)
C      PAUSE'PAUSE BEFORE THE FIRST P(N) CALC IN DPDG'
      P(1) = P2
      DO 20 N = 2,NMAX
       P(N) = P(N-1) - DENS(N)*(Z(N) - Z(N-1))
C     WRITE(0,10) P(N),Z(N),DENS(N)
 20   CONTINUE
C     PAUSE'PAUSE BEFORE THE FIRST CALL CLCVL IN DPDG'
      CALL CLCVL(VCALC)
 25   DO 30 N = 2,NMAX
         PO(N) = P(N)
 30   CONTINUE
C  PAUSE'PAUSE BEFORE THE SECOND P(N) CALC IN DPDG'
      DO 40 N = 2,NMAX
       P(N) = P(N-1) - DENS(N)*(Z(N) - Z(N-1))
C      WRITE(0,10) P(N),Z(N),DENS(N)
 40   CONTINUE
C     PAUSE'PAUSE BEFORE THE SECOND CALL CLCVL'
      CALL CLCVL(VCALC)
      G = PO(NMAX) - P(NMAX)
      AG = ABS(G)
      IF(AG .GT. CHECKP) GO TO 25
      RETURN
      END
C-------------------------------------------------------------------
      SUBROUTINE CLCVL(VCALC)
      COMMON A,B,NMAX,NMAX1,CHECKV,DNSTY,IMAX, IMAX1, PSTMIN,
     *  PSTMAX, EBAR, VE(101), Z(101), ZI(101), ZO(101), SMASS(101),
     *     P(101),PO(101),DENS(101),E(201),CHECKP,DP,XMASS(101),
     *  VETIS(201),VMV(101),VLLUNG(301),VEO(51),VETV(51),
     *  CVG3,FI(51),PMIN,TVE,XAZMAX,ZMAX,SA(101),TSMASS(101),
     *  BLMASS(101),H(101),PCAP(101),HRTPOS,HO,PSI,
     *     PART,K,ZMAX2,FC(2,201)

      VCALC = 0.0
      DO 301 N = 2,NMAX
         II = (P(N) - PSTMIN)/DP + 1
         IF(II .LT. 2) GO TO 100
         IF(II. GT. IMAX1) GO TO 200
         U = P(N) - (PSTMIN + DP*(II-1))
        VE(N) = VETIS(II-1) + U*(VETIS(II + 1) - VETIS(II-1))/(2.*DP)
     &          + U**2*(VETIS(II+1) - 2.0*VETIS(II) + VETIS(II-1))/DP**2
         GO TO 300
 100     U = P(N) - PSTMIN
         VE(N) = VETIS(1) + U*(-1.5*VETIS(1) + 2.0*VETIS(2) -
     &          0.5*VETIS(3))/DP
         IF(P(N) .LE. PSTMIN) VE(N) = VETIS(1)
         GO TO 300
 200    U = P(N) - (PSTMAX -DP)     ! P at IMAX1 =PSTMAX-DP
       . VE(N) = VETIS(IMAX1) + U*(VETIS(IMAX) - VETIS(IMAX1))/DP
 300     IF(VE(N) .LT. 1.00) VE(N) = 1.00
         VCALC = VCALC + VE(N)*VMV(N)
 301  CONTINUE
```

Program REGPVCUR.FOR

```
C
C     CALCULATE THE BLOOD VOLUME, BASED ON THE OLD Z POSITION,
C     AND THE DENSITY, BASED ON THE BLOOD MASS AND THE TISSUE EXPANSION
C
 99   FORMAT(2X,I5,6(1X,F8.2))
      JCOUNT=0
      SUM=0
      VE(1) = VE(2)
      DO 310 N = 2,NMAX
       JCOUNT=JCOUNT+1
         PALV = 0.0
         IF(P(N) .LT. 0.0) PALV = -P(N)
         PCAP(N) = PART + ((Z(N) + Z(N-1))/2.0 - HRTPOS)
         PTM = PCAP(N) - PALV
         H(N) = H0 + PSI*PTM
         IF(PTM .LT. 0.0) H(N) = 0.0
         SMASS(N) = TSMASS(N) + BLMASS(N)
       DENS(N) = SMASS(N)/(VMV(N)*(VE(N) + VE(N-1))/2.0 + BLMASS(N))
       SUM=SUM+VE(N)*VMV(N)
C      WRITE(0,99) N,DENS(N),SMASS(N),VMV(N),VE(N),TSMASS(N),SUM
       IF(JCOUNT.LT.20) GOTO 310
C      PAUSE'PAUSE TO LOOK AT N,DENS,SMASS,VMV,VE,TSMASS,CALCVOL'
      JCOUNT=0
 310  CONTINUE

      DO 400 N = 2,NMAX
C
C     CALCULATE THE NEW Z POSITIONS
C
         IF (Z(N-1) .GT. ZMAX2) GO TO 350
         G1 = 1.0
         G2 = -CVG3
         G3 = 0.0
         Z1 = ZO(N)
 315     G4 = CVG3*Z(N-1)**2 - Z(N-1)**3 + 3.0*SMASS(N)/(DENS(N)*TVE)
 320   CALL SOLVEZ(G1,G2,G3,G4,Z1,Z(N))
         IF(Z(N) .GT. 0.0) GO TO 400
         Z1 = Z1 + 1.0
         GO TO 320
 350     Z(N) = Z(N-1) + SMASS(N)/(DENS(N)*XAZMAX)
 400  CONTINUE
      DO 500 N = 2,NMAX
         ZO(N) = Z(N)
 500  CONTINUE
      RETURN
      END
C-----------------------------------------------------------------------
      SUBROUTINE VRYMNV(FCMIN,TLC,GRAMS,IM,AMV,JSTATUS)

       COMMON A,B,NMAX,NMAX1,CHECKV,DNSTY,IMAX, IMAX1, PSTMIN,
     *  PSTMAX, EBAR, VE(101), Z(101), ZI(101), ZO(101), SMASS(101),
     *     P(101),PO(101),DENS(101),E(201),CHECKP,DP,XMASS(101),
     *  VETIS(201),VMV(101),VLLUNG(301),VEO(51),VETV(51),
     *  CVG3,FI(51),PMIN,TVE,XAZMAX,ZMAX,SA(101),TSMASS(101),
     *  BLMASS(101),H(101),PCAP(101),HRTPOS,H0,PSI,
     *     PART,K,ZMAX2,FC(2,201)

       INTEGER CRT, OFILE
       COMMON /DEVS/ CRT, LUIN, OFILE, LUPLT

C
C  DETERMINE THE PRESSURE AT WHICH THE VOLUME STARTS TO CHANGE -
C  PSTMIN; AND CHANGE THE FC DATA TO VOLUME EXPANSION (VETIS) DATA
```

Program REGPVCUR.FOR

```
C
  3   FORMAT(5X,I5,4F12.4)
  4   FORMAT(10X,I3,4F12.3)
      DPC = (PSTMAX - PMIN)/(IM-1)
      I = 1
 5    I = I+1
      IF(FC(Jstatus,I) .GT. FC(Jstatus,1)) GO TO 6
      GO TO 5
 6    PSTMIN = PMIN + (I-2)*DPC
C     WRITE(0,3)IM,PSTMIN,PMIN,PSTMAX,DPC
C     PAUSE'PAUSE AFTER PRINTING OF IM,PSTMIN,PMIN,PSTMAX,DPC'
      ILF = I-2
      IMF = IM - ILF
      DO 7 I = 1,IMF
           IXF = I + ILF
         FC(Jstatus,I) = FC(Jstatus,IXF)
 7      CONTINUE
      DP = (PSTMAX - PSTMIN)/IMAX1
      FVCS = FC(Jstatus,IM) - FCMIN
      FVCD = FC(Jstatus,IM) - FC(Jstatus,1)
      SCALE = FVCS/FVCD
      E(1) = FCMIN
      DO 8 I = 2,IMF
         E(I) = E(I-1) + SCALE*(FC(Jstatus,I) - FC(Jstatus,I-1))
 8      CONTINUE
      DO 9 I = 1,IMF
         FC(Jstatus,I) = E(I)
 9      CONTINUE
      DO 10 I = 1,IMF
         VETIS(I) = (FC(Jstatus,I)*TLC + GRAMS)/AMV
 10   CONTINUE
C     JCOUNT=0
C     DO 15 I = 1,IMF
C        JCOUNT=JCOUNT+1
C        PR = PSTMIN + (I-1)*DPC
C        DENSC=GRAMS/(FC(Jstatus,I)*TLC+GRAMS)
C        WRITE(0,4) I, PR, VETIS(I), DENSC, FC(Jstatus,I)
C        IF(JCOUNT.LT.20) GO TO 15
C        PAUSE'PAUSE TO LOOK AT I, PR, VETIS(I), DENSC, FC(Jstatus,I)'
C        JCOUNT=0
 15   CONTINUE
C     PAUSE 'PAUSE TO LOOK AT I, PR, VETIS(I), DENSC, FC(Jstatus,I)'
      CALL INTERP(VETIS,IMF,DPC,IMAX,DP)
           JCOUNT=0
C     DO 20 I = 1,IMAX
C        JCOUNT=JCOUNT+1
C        PR = PSTMIN + (I-1)*DP
C        DENSC=GRAMS/(FC(Jstatus,I)*TLC+GRAMS)
C        WRITE(0,4) I, PR, VETIS(I), DENSC, FC(Jstatus,I)
C        IF(JCOUNT.LT.20) GO TO 20
C        PAUSE'PAUSE TO LOOK AT I, PR, VETIS(Jstatus,I), DENSC, FC(Jstatus,I)'
C        JCOUNT=0
C 20  CONTINUE
      RETURN
      END
C-------------------------------------------------------------------
        SUBROUTINE INTERP(Y,IMAX1,DX1,IMAX2,DX2)
      DIMENSIONY(211), X1(211), X2(211), Y2(211)
      DO 10 I1 = 1,IMAX1
           X1(I1) = (I1 -1)*DX1
   10   CONTINUE
      DO 20 I2 = 1,IMAX2
           X2(I2) = (I2-1)*DX2
   20   CONTINUE
      DO 30 I2 = 1,IMAX2
```

```
          I1 = X2(I2)/DX1 + DX1/2.0
          I1 = I1 +1
          IF(I1 .EQ. 1) I1 = 2
          IF(I1 .EQ. IMAX1) I1 = IMAX1 - 1
          U = (X2(I2) - X1(I1))/DX1
          D1 = Y(I1+1) - Y(I1-1)
          D2 = Y(I1+1) - 2.0*Y(I1) + Y(I1-1)
          Y2(I2) = Y(I1) + 0.5*D1*U + 0.5*D2*U**2
   30   CONTINUE
      DO 40 I2 = 1,IMAX2
          Y(I2) = Y2(I2)
   40   CONTINUE
      RETURN
        END
C-----------------------------------------------------------------
      SUBROUTINE SOLVEZ(A,B,C,D,Z0,Z2)
      Z2 = Z0
 10   F = A*Z2**3 + B*Z2**2 + C*Z2 + D
      AF = ABS(F)
        IF(AF .GE. 0.05) THEN
          FP = 2.0*A*Z2**2 + 2.0*B*Z2 + C
          Z2 = Z2 - F/FP
          GO TO 10
        END IF
      RETURN
      END
```

Appendix C - 3
Ventilation Model

PROGRAM SBBOLUSD

```
DIMENSION DV(7),CBOLUS(30),VMAX(10),CBOLreg(10)
       COMMON FLOW(601),VOLIN(601),F(15),RTLC(15),WL(25),WD(25),
     + P1(201),V(7),AWL(7,18),H(7),Z(10),Z0(10),RCV(7),DVdt,COEFA,
     + AWD(7,18),Q(7),Pst(2,7,151),PTM(7,30),DENSITY,VISCSTY,VISK,
     + P(15),DPR(15),RES(15),DPRL(15),DMAX(25),VO(15),JCOUNT,U(7,20),
     + VOLD(15),K,ImaxFC,NUNITS,DFC,FCMIN,APGPL(25),NGEN,DD,XLMAX(24),
     + PERIOD,DT,TMAX,FCMAX,DELVOL,FDMAX(201),XKM1(7),XK(7,25),
     + DPG(7,18),RESCON(7,18),RESVIS(7,18),JINSIGN,JEXSIGN,FCFRC,TLC,
     + DPptm,ImaxFCM1,JCYCLE,PTMMIN,PTMMAX,VE,VMINLUNG,WEIGHT,DPGRAV(7),
     + FVt(201),ImaxFCP1,Jmax,TXAR(7,30),POIS(7,30),FDMAX0,SLOPE0,
     + JSTADYN,JQSTATIC,DPGRAV0(7),DFCS,DFC2,AMOUNT(7),GI,JmaxPTM,
     + NGENDP,NGENDP1,DK,QG(5),RWEIGHT(10),RVE(10),RMINVOL(10),QR(10),
     + DPV(7,30),DPC(7,30)
C
C  COMMON STATEMENTS FOR IODECLS
C
      INTEGER A1, A2
      INTEGER CRT, LUOUT, LUIN, LUPLT
      CHARACTER*1 BEEP
      CHARACTER*80 ITITLE
      CHARACTER*64 NAINP,NAOUT,NAPLT
      COMMON /DEVS/LUTRM, LUIN, LUOUT, LUPLT
      COMMON /CURSOR/ BEEP
      COMMON /LABELS/ITITLE,NAINP,NAOUT,NAPLT
C
 1    FORMAT(4F10.3)
 2    FORMAT(5F10.4,I10)
 3    FORMAT(13F6.3)
 4    FORMAT(I3,2F10.4)
 5    FORMAT(13F6.2)
 6    FORMAT(10E13.3)
 7    FORMAT(4I5,F10.2)
 9    FORMAT(/I10,3F15.3)
 10   FORMAT(4F25.5)
 14   FORMAT(2X,I10,7F14.5)
 15   FORMAT(2I10,8F12.4)
 18   FORMAT(2E15.3)
 20   FORMAT('  J', 9X,'Z(J)', 9X,' Q(J) ', 9X,' V(J) ', 9X,' H ',
     +9X,'  PVC ', 9X,'   FTV ', 9X,' P(J) ',8X,'DPRL(J)',7X,'RCF(J)')
 21   FORMAT(I3,F15.3,2F15.2,5F15.4,F13.4)
 23   FORMAT(1H1)
 25   FORMAT(/)
 26   FORMAT(10F8.1)
 28   FORMAT(10F8.4)
 29   FORMAT(1X,9F8.4)
 30   FORMAT(1X,I5,2F10.2)
 31   FORMAT(1X,I4,F8.1,4F8.2,4F8.1)
 32   FORMAT(2X,I5,5F9.2)
 33   FORMAT(I5,F6.1)
 34   FORMAT(5F11.5)
 35   FORMAT(10F8.1)
       CRT=0
C
C          DEFINE THE TYPE OF BREATH THAT WILL BE ANALYZED
C            JSTADYN             INHALATION  EXHALATION
C               1                 DYNAMIC    DYNAMIC
C               2                 STATIC    STATIC
C               3                 DYNAMIC    STATIC
C               4                 STATIC    DYNAMIC
C
```

Program SBBOLUSD.FOR

```
      JSTADYN=1
      DVdt=20           !Volume increment per DT time step
C
C  READ IN THE LUNG VOLUMES FOR THIS SUBJECT FROM THE DATA FILE CREATED
C  IN REGPFCUR.  ALPHA AND BETA ARE EXPANSION COEFFICIENTS AND
C  VMINLUNG IS THE MINIMUM  VOLUME OF THE LUNGS (GAS AND PARECHYMA),
C  AND WEIGHT IS THE TOTAL MASS OF (HE LUNGS.
C
      CALL IODECLS(1,0,0)
      READ(LUIN,2) RV,FRC,TLC,DEADSP
      READ(LUIN,2) ALPHA,BETA,WEIGHT
      READ(LUIN,2) (DPGRAVO(I),I=4,6)
C
C    READ IN THE LUNG FUNCTIONAL DATA.IT IS ASSUMED THAT THE LUNG
C    TISSUE HAS NO HYSTERESIS AND ALL REGIONS HAVE THE SAME TISSUE
C    CHARACTERISTICS.  THE REGIONALP P-V CURVES HAVE BEEN CALCULATED
C    OUTSIDE THIS PROGRAM WITH THESE ASSUMPTIONS.
C
      READ(LUIN,4) ImaxFC,FCMIN,FCMAX
      DO 40 J=4,7
      DO 40 Jstatus = 1,2
  40  READ(LUIN,28) (Pst(Jstatus,J,I),I=1,ImaxFC)

      DFC=(FCMAX-FCMIN)/(ImaxFC-1)
      DFC2=2.0*DFC
      DFCS=DFC**2
      SLOPE=5.0/DFC
      DO 41 J=4,7
      DO 41 Jstatus=1,2
C ADD TWO LINEAR POINTS TO THE BEGINNING OF THE CURVE
      DO 39 I=ImaxFC,1,-1
 39   Pst(Jstatus,J,I+2)=Pst(Jstatus,J,I)
      Pst(Jstatus,J,2)=Pst(Jstatus,J,3)-SLOPE*DFC
      Pst(Jstatus,J,1)=Pst(Jstatus,J,3)-2.0*SLOPE*DFC
C ADD TWO LINEAR POINTS ONTO THE END OF THE CURVE

      Pst(Jstatus,J,ImaxFC+2)=Pst(Jstatus,J,ImaxFC+1)+SLOPE*DFC
      Pst(Jstatus,J,ImaxFC+3)=Pst(Jstatus,J,ImaxFC+1)+2*SLOPE*DFC
  41  Pst(Jstatus,J,ImaxFC+4)=Pst(Jstatus,J,ImaxFC+1)+3*SLOPE*DFC

      ImaxFC=ImaxFC+4
      ImaxFCM1=ImaxFC-1
      ImaxFCP1=ImaxFC+1

      FCMIN=FCMIN-2.0*DFC
      FCMAX=FCMAX+2.0*DFC

C
C          READ IN THE EXHALATION PARENCHYMAL P-FV RELATION
C                    FOR CALCULATING AIRWAY DIAM

      READ(LUIN,4) Jmax,PTMMIN,PTMMAX
      READ(LUIN,28)(FVt(J),J=1,Jmax)

C   ADD HYSTERESIS TO MAKE P-FV CURVE AN INHALATION CURVE
C        HYSTERESIS IN A VC BREATH IS 4.0cm H2O

      DPHYS=4.0
      PTMMIN=PTMMIN+DPHYS
      PTMMAX=PTMMAX+DPHYS
      .
C     WRITE(CRT,4) Jmax,PTMMIN,PTMMAX
C     WRITE(CRT,28)(FVt(J),J=1,Jmax)
C     PAUSE'PAUSE TO LOOK AT IMFVA,PTMMIN,PTMMAX AND FVt(I),I=1,IMFVA'
```

Program SBBOLUSD.FOR

```
      JmaxM1=Jmax-1

C  DEADSPACE=160cc     TOTAL ANATOMICAL DEADSPACE
C  RDS=25cc                ANATOMICAL DEADSPACE WITHIN A REGION THAT REPRESENTS
C                    A QUARTER OF THE TOTAL LUNG MASS
C  UAWDS=60cc              ANATOMICAL DEADSPACE FOR THE UPPER AIRWAYS
C                    (MOUTH,THROAT,TRACHEA)

      DEADSPACE=160.0
      RDS=25.0
      UAWDS=60.0

      TLCT=TLC
      FRCT=FRC
      TLC=TLC-DEADSP
      FRC=FRC-DEADSP
      RV=RV-DEADSP
      FCFRC=FRC/TLC
      VMINLUNG=FCMIN*TLC+WEIGHT
C
C  OPEN OUTPUT FILE
C
       CALL IODECLS(0,1,0)
C
C   INPUT DATA FOR WEIBEL'S LUNG MODEL A FROM WEIBEL.DAT
C
       NI5=5
C   OPEN INPUT FILE 'WEIBEL.DAT'
      OPEN(NI5,FILE='WEIBEL.DAT')
      NGEN=23                       !Number of airway generations
      READ(NI5,1) VOM               !Total gas volume
      READ(NI5,3) (WL(N),N=1,23)     !Airway lengths in cms
      READ(NI5,3) (WD(N),N=1,23)     !Airway diameters in cms
      WTRDIAM = 1.8                  !Trachea diameter in cms

C   READ IN THE BOLUS CONCENTRATION AT THE END OF THE TRACHEA
C   THIS PROFILE WAS CALCULATED OUTSIDE THIS PROGRAM

       NI4=4
       OPEN(NI4,FILE='CBOLUS.DAT')
       READ(NI4,33) JmaxB,DVbolus
       READ(NI4,34) (CBOLUS(JB),JB=1,JmaxB)
       JmaxBP1=JmaxB+1
C     WRITE(CRT,33) JmaxB,DVbolus
C     WRITE(CRT,34) (CBOLUS(JB),JB=1,JmaxB)
C     PAUSE'LOOK AT INPUT DATA'

C
C  SUBSCRIBTS THAT DEFINE THE DIFFERENT AREAS OF THE LUNG MODEL
C          SUBSCRIBT 1 - ENTIRE LUNG
C          SUBSCRIBT 2 - UNIT COMPOSED OF UPPER TWO REGIONS 1 and 2
C          SUBSCRIBT 3 - UNIT COMPOSED OF LOWER TWO REGIONS 3 and 4
C          SUBSCRIBT 4 - REGION 1
C          SUBSCRIBT 5 - REGION 2
C          SUBSCRIBT 6 - REGION 3
C          SUBSCRIBT 7 - REGION 4
C
      F(1)=1.0            !F(I) IS THE LUNG MASS OF EACH AREA
      F(2)=.5
      F(3)=.5
      F(4)=.25
      F(5)=.25
      F(6)=.25
      F(7)=.25
C
```

Program SBBOLUSD.FOR

```
C     INITIALIZE VARIABLES
C
      DO 36 J=1,7
      AMOUNT(J)=0.0
      RTLC(J)=F(J)*TLC          !Regional TLC, does not include the upper
      RWEIGHT(J)=F(J)*WEIGHT
      RMINVOL(J)=F(J)*VMINLUNG
      RCV(J)=.5
 36   DPRL(J)=.1

      DPFG=0.0
      FDG=1.0
      IBOLUS=1
      C=0.0

C
C  CHECK INPUT DATA
C
C     WRITE(CRT,10) FRC,TLC
C     PAUSE'PAUSE TO LOOK AT FRC,TLC  '
C     WRITE(CRT,28) ( F(J),J=1,7)
C     PAUSE'PAUSE TO LOOK AT F(J) '
C     WRITE(CRT,26) ( RTLC(J),J=1,7)
C     PAUSE'PAUSE TO LOOK AT RTLC(J) '
C     WRITE(CRT,3) (WL(N),N=1,23)
C     PAUSE'PAUSE TO LOOK AT  WL'
C     WRITE(CRT,3) (WD(N),N=1,23)
C     PAUSE'PAUSE TO LOOK AT  WD'
      DO 44 J=4,7
      DO 44 Jstatus=1,2
      WRITE(CRT,29) (Pst(Jstatus,J,I),I=1,ImaxFC)
      PAUSE'PAUSE TO LOOK AT PST '
 44   CONTINUE
C     WRITE(CRT,29) (FVt(J),J=1,Jmax)
C     PAUSE'PAUSE TO LOOK AT THE FVA'
C
C     READ IN THE EXPERIMENTAL CONDITIONS

      DENSITY =0.001121              !gms/cm**3
      VISCSTY =0.0001914             !gm/(sec*cm)
      VISK=VISCSTY/DENSITY           !Kinematic VIScosity (cm**2/sec)
      GI=1/980.2                     !dynes/cm**2=(1/980.2)cmH20
      DD=GI*DENSITY/2
      DK=DENSITY*0.85               !0.85 experimental constant for conv acc
      EXC=1.85                      !1.85 experimental constant for vis losses
      COEFA=EXC*5.659*VISCSTY       !COEFficient for Vis Losses
      NGENDP=16                     !# of gen used in DP calc
      NGENDP1=NGENDP+1
C
C     WRITE(CRT,6) VISCSTY,DENSITY,VISK
C     PAUSE'PAUSE TO LOOK AT VISCSTY, DENSITY, VISK'
C
      CALL FLOWW(KMAX,KEINS,VSTART)
      V(1)=VSTART-DEADSP        !gas Volume of area 1 at the start
C
C  USE CONSTANT VOLUME INCREMENTS OF DVdt SIZE IF THIS IS A STATIC BREATH
C
      IF(JSTADYN.NE.2) GOTO 49
      VOLIN(1)=0.0
      DO 47 K=2,KEINS
 47   VOLIN(K)=VOLIN(K-1)+DVdt
      KEINSP1=KEINS+1
      DO 48 K=KEINSP1,KMAX+1
 48   VOLIN(K)=VOLIN(K-1)-DVdt
 49   CONTINUE
```

Program SBBOLUSD.FOR

```
C
C     WRITE(CRT,2) TIDVOL, DT, TMAX
C      PAUSE'PAUSE TO LOOK AT TIDVOL, V(1), DT, TMAX '
C
C  CALCULATE THE MAXIMUM LENGTH AND DIAMETER OF EACH GENERATION
C
      SCALE=(TLCT/VOM)**(1./3.)      !The linear scaling of VOM to TLCT
      TRCDMAX=WTRDIAM*SCALE          !Maximum trachea diameter
      DO 50 N=1,NGEN
      XLMAX(N)=SCALE*WL(N)           !Maximum airway lengths
   50 DMAX(N)=SCALE*WD(N)            !Maximum airway diameters

C     WRITE(CRT,3) (XLMAX(N),N=1,NGEN)
C      PAUSE'PAUSE TO LOOK AT (XLMAX(N),N=1,NGEN)'
C     WRITE(CRT,3) (DMAX(N),N=1,NGEN)
C      PAUSE'PAUSE TO LOOK AT (DMAX(N),N=1,NGEN)'
C
C          CALCULATION OF OMEGA - XKM=2.0 AT P=5CMH2O
      I=1
   54 IF(Pst(Jstatus,6,I).GE.5.0) GO TO 56
      I=I+1
      GO TO 54
   56 OMEGA=(Pst(Jstatus,6,I+1)-Pst(Jstatus,6,I-1))/(2.*DFC)
C     WRITE(CRT,1) OMEGA        !NOTE OMEGA=19.0 cm H2O CHECK!!
C     PAUSE'PAUSE TO LOOK AT OMEGA'
C
      XKB=2.474          !2.474 cm H2O (3.21 and 3.22)
C
C   CALCULATION OF D/DMAX-PTM RELATION FOR POSITIVE PTM.  THE
C   AIRWAYS ARE ASSUMED TO EXPAND UNIFORMLY WITH THE PARENCHYMA
C   AND THE TRANSMURAL PRESSURE IS EQUAL TO THE TRANSPULMONARY PRESSURE.

      DPptm=(PTMMAX-PTMMIN)/(JMAX-1)
      PTMlow=PTMMIN-DPHYS          !Use the exhalation P-FVt curve

C  DETERMINE AT WHAT J VALUE THE PTM BECOME POSITIVE
      J=0
   58 J=J+1
      Pptm = PTMlow+(J-1)*DPptm
      IF(Pptm.LT.0.0) GOTO 58
      Jstart=J-1
      IF(Jstart.EQ.0) Jstart=1
      JmaxPTM=JMAX-(Jstart-1)

      JJ=0
      DO 60 J=Jstart,Jmax
      JJ=JJ+1
   60 FDMAX(JJ)=FVt(J)**(1./3.)
      WRITE(CRT,29) (FDMAX(J),J=1,JmaxPTM)
      PAUSE'PAUSE TO LOOK AT FDMAX(J),J=1,JmaxPTM'

C  CALCULATE PARAMETERS NEEDED FOR THE TUBE LAW (NEGATIVE PTM's)
      FDMAX0=FDMAX(1)
      SLOPE0=DPptm/(FDMAX(2)-FDMAX(1))

      CHECKP=.001          !Pressure error in iteration, cmH2O
      DPHYS=0.00           !Hysteresis for a VC breath
C
C    CALCULATE THE REGIONAL VOLUMES, TRANSMURAL PRESSURES
C    AND AIRWAY DIAMETERS BEFORE THE BREATH STARTS.
C
      VELUNG=(V(1)+WEIGHT)/VMINLUNG
      FCLUNG=V(1)/TLC
C     WRITE(CRT,29) V(1),WEIGHT,VMINLUNG,TLC
C     PAUSE'PAUSE TO LOOK AT V(1),WEIGHT,VMINLUNG,TLC'
```

Program SBBOLUSD.FOR

```
      DO 63 J=4,6
   63 DPGRAV(J)=DPGRAVO(J)/VELUNG**(ALPHA+BETA)
C     WRITE(CRT,6)V(1),WEIGHT,VMINLUNG,FCLUNG,(DPGRAV(J),J=4,6)
C     PAUSE' PAUSE; V(1),WEIGHT,VMINLUNG,FCLUNG,DPGRAV(J),J=4,6 '

      K=1    !K=1 IS THE LUNG VOLUME AT STRART OF BREATH, TIME=0
      T=0.0                        !TIME ZERO IS THE START OF THE BREATH
C
      Jstatus=1                    !Inhalation
      Iprint=1
      CALL STATVOL(FClung,Jstatus)
      CALL TRANMPR(OMEGA,DPFG)
      CALL AIRWAY(TRCDMAX,TRCDIAM)
C
C-----------------------------------------------------------------
C   CALCULATIONS OF THE REGIONAL FLOWS AT EACH DT TIME INCREMENT
C          INITIALIZE PARAMETERS
C
      JCYCLE=1            !JCYCLE=1 IS INHALATION, =-1 IS EXHALATION
      JINSIGN=1
      JEXSIGN=-1
C  Z DEFINES THE AMOUNT OF FLOW GOING TO EACH OF THE AREAS OF THE MODEL
C
      Z(1)=1.0       !Z defines the fraction of flow going to
      Z(2)=0.5       !  a region, Z(Region)
      Z(3)=0.5
      Z(4)=0.25
      Z(5)=.25
      Z(6)=.25
      Z(7)=.25
C      PAUSE 'PAUSE AFTER THE Zs ARE DEFINED'
C
C     AT T = 0 THE SYSTEM IS IN ELASTIC EQUILIBRIUM.
C     THE DIFFERENCES IN QU AND QL IN EACH SET IS DUE TO
C     THE DIFFERENT AIRWAY RESISTANCES
C
      K=1     !K=1 IS THE INITIAL TIME STEP (TIME=0)
      T=0.0             !TIME ZERO IS THE START OF THE BREATH

      WRITE(LUOUT,31)K,V(1),(P(J),J=4,7),(V(J),J=4,7)
      WRITE(LUOUT,2499)
C----------------------------------------------------------------
 1000 T=T+DT
      K=K+1
      IF(K.GT.KMAX) GOTO 5000
      DO 1002 J=1,7
 1002 VO(J)= V(J)
      IF(K.LT.257) GOTO 1003
      KKK=1
 1003 CONTINUE
      Q(1)=FLOW(K)      ! Q(1) IS FLOW IN THE TRACHEA-TOTAL FLOW
      AQ=ABS(Q(1))
      DELVOL=VOLIN(K)-VOLIN(K-1)
      V(1)=V(1)+DELVOL
           WRITE(CRT,32) K,Q(1),DELVOL,V(1)

C        PAUSE'PAUSE TO SEE K,Q(1),DELVOL AND V(1) FOR THE NEXT TIME STEP'
C
C  CALCULATE THE VOLUME EXPANSION OF THE LUNG AND THE FOUR REGIONS
C  AND THE GRAVITATIONAL EFFECT ON THE REGIONAL PARENCHYMAL PRESSURES
C
      VE=(V(1)+WEIGHT)/VMINLUNG
      DO 1005 R=4,6
      VEXP=(RVE(R)+RVE(R+1))/2.0
 1005 DPGRAV(R)=DPGRAVO(R)/VEXP**(ALPHA+BETA)
```

Program SBBOLUSD.FOR

```
C       WRITE(CRT,29) (DPGRAV(R),R=4,6)
C       PAUSE'PAUSE TO LOOK AT THE DPRGAV(R),R=4,6'
      IF (K.EQ.KEINS) GO TO 3000
 1006 CONTINUE        ! return after setting exhalation values
      IF(JSTADYN.EQ.2) GO TO 1010
C       WRITE(CRT,14) K,Q(1),DELVOL
C       PAUSE'PAUSE TO LOOK AT K, Q(1), DELVOL'
      IF(AQ.LT.50.0) GO TO 1010
      GO TO 1040
 1010 JQSTATIC=2
C
C     JQSTATIC=2 INDICATES QUASISTATIC FLOW,FLOWS < 50 cc/sec.
C          USE STATVOL TO CALCULATE THE FLOW RATES
C
      DELVOL=ABS(DELVOL)
      FCLUNG=V(1)/TLC
      CALL STATVOL(FClung,Jstatus)
      DO 1020 J=2,7
      DPRL(J)=0.0
 1020 RCV(J)=0.0
      GO TO 1405
 1040 CONTINUE
      JQSTATIC=1
C
C     JQSTATIC=1 INDICATES DYNAMIC FLOW (FLOWS > 50 cc/sec).
C
C   TO DETERMINE THE REGIONAL FLOWS ITERATE ABOUT THE FLOW TO
C   AREA 2 (Q(2)).  NEED TWO GUESSES FOR Q(2) TO GET THE ITERATION
C   STARTED.  THE FIRST GUESS Q1 IS BASED ON THE FLOWS AT THE
C   PREVIOUS TIME STEP. Q2 AND Q3 ARE THE LATEST GUESSES FOR Q(2).
C   THE ITERATION FUNCTION F IS THE ERROR WHEN THE PRESSURES IN EQ.
C   3.32 ARE NOT PROPERLY BALANCED.  ONCE THE ERROR IS BELOW CHECKP
C   THE ITERATION IS STOPPED.
C
      Q2=Z(2)*Q(1)          ! first quess for Q(2)
      QG(2)=Q2                ! define QG(2) and QG(3) for use in
      QG(3)=Q(1)-QG(2)        !    REGVOL
C       WRITE(CRT,29) Z(2),Q(1),Q2
C       PAUSE'PAUSE TO LOOK AT Z(2),Q(1),Q2 IN MAIN'
C
      A1=4    !beginning Area for PRESSLS calculations
      A2=7    !end Area for PRESSLS calculations
      LSIT=1           !LSITuation =1, 1st time enter PRESSLS in K time step
      CALL PRESSLS(A1,A2,LSIT,Q2)
      CALL REGFLOW(CHECKP)
C
C          LSIT =2 FOR ALL OTHER ENTRIES INTO PRESSLS
C
C     DPRL IS THE PRESSURE DROP FROM THE ALVEOLI IN THE ITH
C     POSITION OF THE TRACHEA
C     THE FUNCTION USED IN THE ITERATION IS THE ERROR WHEN THE
C     PRESSURES BETWEEN AREAS 2 AND 4 ARE NOT BALANCED.
C
      IGUESS=1
      DPRL(5)=DPR(5)+DPG(2,2)
      DPRL(7)=DPR(7)+DPG(3,2)
      G=P(5)-P(7)-(DPGRAV(5)+DPGRAV(6))
      F2=G+(DPRL(7)-DPRL(5))
      AF2=ABS(F2)
      IF(K.LT.71) GOTO 1290
C     WRITE(LUOUT,7) K,LSIT
C     WRITE(LUOUT,32) IGUESS,F2,DPRL(5),DPRL(7),DPG(2,2),DPG(3,2)
 1290 CONTINUE
C       WRITE(CRT,29) G,P(5),P(7),DPGRAV(5),DPGRAV(6),DPRL(5),DPRL(7)
C       PAUSE'PAUSE TO LOOK AT G,P(5),P(7),DPGRAV(5),DPGRAV(6),DPRL(5)
```

```
C     +      ,DPRL(7), READY TO GET SECOND GUESS FOR Q(2) IN MAIN'
      Q3=Q2              !Define Q3 in case AF2.LT.CHECKP
      IF(AF2.LT.CHECKP) GO TO 1400
C                                       SECOND GUESS
      IF(F2.GT.0.0) Q3=1.02*Q2
      IF(F2.LT.0.0) Q3=0.98*Q2
C
 1300 F1=F2
      Q1=Q2
      Q2=Q3
      IGUESS=IGUESS+1
C
      QG(2)=Q2  !define QG(2) and QG(3) for use in REGVOL calc
      QG(3)=Q(1)-QG(2)
C        WRITE(CRT,29) Q2,QG(3)
C        PAUSE'PAUSE TO LOOK AT THE NEXT GUESS FOR Q2 AND Q3 IN MAIN'
      LSIT=2
      A1=4
      A2=7
      CALL PRESSLS(A1,A2,LSIT,Q2)
      CALL REGFLOW(CHECKP)
      DPRL(5)=DPR(5)+DPG(2,2)
      DPRL(7)=DPR(7)+DPG(3,2)
      F2=G+(DPRL(7)-DPRL(5))
      AF2=ABS(F2)
C        WRITE(CRT,15) IJUMP
C        WRITE(CRT,29) Q1,Q2,F1,F2,DPRL(5),DPRL(7)
C        PAUSE'PAUSE TO LOOK AT LCOUNT,Q1,Q2,F1,F2 IN MAIN'
      IF(AF2.LT.CHECKP) GO TO 1400
C                                       NEXT GUESS
      Q3=(Q1*F2-Q2*F1)/(F2-F1)
      IF(K.LT.71) GOTO 1390
C     WRITE(LUOUT,7) K,LSIT
C     WRITE(LUOUT,32) IGUESS,F2,DPRL(5),DPRL(7),DPG(2,2),DPG(3,2)
 1390 CONTINUE
      IF(IGUESS.GT.25) GO TO 5000
      GO TO 1300
 1400  CONTINUE
C
C       ITERATION FOR Kth TIME STEP COMPLETE
C
      Q(2)=Q3
      Q(3)=Q(1)-Q(2)
      DO 1402 J=1,7
 1402 ZO(J)=Z(J)

      GOTO 1420

 1405 CONTINUE

C             CALCULATE THE Zs FOR THIS TIME STEP

C                      QUASISTATIC CASE

      DO 1407 J=4,7
      DV(J)=V(J)-VO(J)
 1407 Z(J)=DV(J)/DELVOL    !CHANGE MADE CHECK OUT!!!!!!!!!!!!!!!!!

      SUM=Z(4)+Z(5)+Z(6)+Z(7)

      DO 1408 J=4,7
 1408 Z(J)=Z(J)/SUM
      IF(JSTADYN.EQ.2) GOTO 2100

      DO 1410 J=4,7
```

Program SBBOLUSD.FOR

```
 1410 Q(J)=DV(J)/DT
      Q(2)=Q(4)+Q(5)
      Q(3)=Q(6)+Q(7)
      GOTO 2100

 1420 CONTINUE
C                            DYNAMIC CASE
      DO 1450 J=1,7
 1450 Z(J)=Q(J)/FLOW(K)
C        WRITE(CRT,14) K
C        WRITE(CRT,29) (Z(J),J=1,7)
C        PAUSE'PAUSE TO LOOK AT THE Zs AT THE END OF THE Kth TIME STEP'
      IF(JSTADYN.EQ.1.AND.JQSTATIC.EQ.1) GO TO 1800
      IF(Q(1).EQ.0.0) GO TO 1800
      SUM=0.0
      DO 1560 J=4,7
 1560 SUM=SUM+Z(J)
      DO 1600 J=4,7
 1600 Z(J)=Z(J)/SUM
 1800 CONTINUE
 1810 VOLD(1)=V(1)
      VOLD(2)=V(2)
      VOLD(3)=V(3)
C
C     INTEGRATE THE FLOW TO EACH REGION (AREA=4,7) AND CALCULATE THE
C     NEW RECOIL PRESSURE OF THE PARENCHYMA OF THAT REGION
C
      DO 2000 J=4,7
      VOLD(J)=V(J)
      FVREG=V(J)/RTLC(J)
      IF(JQSTATIC.EQ.2) GO TO 1980
      IF(JSTADYN.EQ.2) GO TO 1980
      DV(J)=DT/2.0*(ZO(J)*FLOW(K-1)+Z(J)*FLOW(K))
      V(J)=V(J)+DV(J)

      I=0
 1820 I=I+1
      IF(I.EQ.ImaxFC) GOTO1840
      FCI=FCMIN+DFC*(I-1)
      IF(FCI.LT.FVREG) GOTO 1820
      IF(I.EQ.1) GO TO 1830

      UU=FVREG-(FCI-DFC)
      P(J)=Pst(Jstatus,J,I-1)
     *        + UU*(Pst(Jstatus,J,I)-Pst(Jstatus,J,I-1))/DFC
      GO TO 1980

 1830 UU=FVREG-FCMIN
      P(J)=Pst(Jstatus,J,1)
     *        + UU*(Pst(Jstatus,J,2)-Pst(Jstatus,J,1))/DFC
      GO TO 1980

 1840 UU=FVREG-FCMAX
      P(J)=Pst(Jstatus,J,ImaxFC)
     *        + UU*(Pst(Jstatus,J,ImaxFC)-Pst(Jstatus,J,ImaxFCM1))/DFC

 1980 CONTINUE
      PVC=(V(J)-VO(J))/VO(J)
      FTV=(V(J)-VO(J))/VOLIN(K)
C        WRITE(CRT,21) J,Q(J),V(J),P(J),DPRL(J)
C        PAUSE'PAUSE TO LOOK AT J,Q(J),V(J),P(J),DPRL(J)'
 2000 CONTINUE
C
C   CALCULATE THE NEW VOLUMES OF AREAS 2 AND 3
C
```

```
      V(2)=V(4)+V(5)
      V(3)=V(6)+V(7)
C
C  CALCULATE THE VELOCITY AND DPs IN THE TRACHEA AND THE 1st GEN
C
      VTRACH=ABS(Q(1)/(.785*TRCDIAM**2))
      COEF=-JCYCLE*0.0107
      DPFR=COEF*U(1,1)*VISCSTY/AWD(1,1)*(U(1,1)*AWL(1,1)/VISK)**0.5
      DPBE=DD*(FDG*VTRACH**2-1.7*U(1,1)**2)
      DPFG=DPFR+DPBE
C
C   CALCULATE THE TRANSMURAL PRESSURES AND THE NEW AIRWAY DIAMETERS
C      AND LENGTHS THAT WILL BE USED IN THE NEXT TIME STEP K+1
C
      CALL TRANMPR(OMEGA,DPFG)
C        PAUSE'PAUSE BEFORE YOU ENTER AIRWAY'
      CALL AIRWAY(TRCDMAX,TRCDIAM)
C
 2100 CONTINUE
      IF(JCYCLE.EQ.-1) GOTO 2200
      IF(IBOLUS.EQ.0) GOTO 2300

C    CALCULATE THE AMOUNT OF INDICATOR MATERIAL THAT ENTERS EACH REGION

       CO=C
       JB=1
 2110  JB=JB+1
       IF(JB.EQ.JmaxBP1) GOTO 2120
       VOLjb=(JB-1)*DVbolus
       IF(VOLIN(K).GT.VOLjb) GOTO 2110
       UU=(VOLIN(K)-VOLJB)/DVbolus
       C=CBOLUS(JB-1)+UU*(CBOLUS(JB)-CBOLUS(JB-1))
       GOTO 2130
 2120  IBOLUS=0
       C=0

 2130  CONTINUE
       DO 2140 J=4,7
 2140  AMOUNT(J)=AMOUNT(J)+DV(J)*(C+CO)/2.0
       GOTO 2300

C   CALCULATE THE GAS CONCENTRATION AT THE MOUTH DURING
C   EXHALATION (CONMO) - EACH REGION HAS RDS OF ANATOMICAL DEADSPACE
C   AND THE UPPER AIRWAYS HAS A DEADSPACE UAWDS

 2200  CONMO=0.0
       DO 2150 J=4,7
       FK=1.0
       RVEXH=VMAX(J)-V(J)
       IF(RVEXH.LT.RDS) FK=0.0
 2150  CONMO=CONMO+FK*Z(J)*CbolREG(J)
       EXHVOL=VMAX(1)-V(1)+UAWDS
       WRITE(LUOUT,30) K,EXHVOL,CONMO
       WRITE(CRT,30) K,EXHVOL,CONMO

 2300  CONTINUE

      Iprint=Iprint+1
      WRITE(LUOUT,31)K,V(1),(P(J),J=4,7),(V(J),J=4,7)
      WRITE(LUOUT,31)K,Q(1),(Z(J),J=4,7),(DPR(J),J=4,7)
      IF(IPRINT.LT.1000) GOTO 2900
      Iprint=0
      DELGRAVITY=DPGRAV(5)+DPGRAV(6)

      IF(JSTADYN.EQ.2) GOTO 2500
```

```
C     CALCULATE PULMONARY RESISTANCE

      DO 2400 J=4,7
 2400 RES(J)=DPR(J)/(Q(J)*.001)

      RESI = DPG(2,2)/(Q(2)*.001) + (RES(4)*RES(5))/(RES(4)+RES(5))
      RESII= DPG(3,2)/(Q(3)*.001) + (RES(6)*RES(7))/(RES(6)+RES(7))

      RESPUL=DPFG/(Q(1)*.001) + (RESI*RESII)/(RESI+RESII)
 2499 FORMAT(/)
 2500 CONTINUE
      WRITE(LUOUT,2499)
      WRITE(LUOUT,2499)
      WRITE(LUOUT,29) DELGRAVITY,DPGRAV(5),DPGRAV(6)
      WRITE(LUOUT,29) DPR(4),DPR(5),DPR(6),DPR(7),DPRL(5),DPRL(7)
      WRITE(LUOUT,2499)
C     WRITE(LUOUT,29) (XKM1(J),J=4,7)
C     WRITE(LUOUT,29) (XK(4,N),N=2,NGENDP)
      WRITE(LUOUT,28) PTM(1,1),PTM(2,2),(PTM(4,N),N=3,NGENDP)
      WRITE(LUOUT,28) AWD(1,1),AWD(2,2),(AWD(4,N),N=3,NGENDP)
      WRITE(LUOUT,28) PTM(1,1),PTM(3,2),(PTM(7,N),N=3,NGENDP)
      WRITE(LUOUT,28) AWD(1,1),AWD(3,2),(AWD(7,N),N=3,NGENDP)
      WRITE(LUOUT,2499)
      WRITE(LUOUT,35)  VTRACH,U(1,1),U(2,2),U(3,2)
      WRITE(LUOUT,35) (U(4,N),N=3,NGENDP)
      WRITE(LUOUT,35) (U(7,N),N=3,NGENDP)
      WRITE(LUOUT,2499)
      WRITE(LUOUT,28) DPFR,DPV(2,2),(DPV(4,N),N=3,NGENDP)
      WRITE(LUOUT,28) DPBE,DPC(2,2),(DPC(4,N),N=3,NGENDP)
      WRITE(LUOUT,28) DPFG,DPG(2,2),(DPG(4,N),N=3,NGENDP)
      WRITE(LUOUT,2499)
      WRITE(LUOUT,28) DPFR,DPV(3,2),(DPV(7,N),N=3,NGENDP)
      WRITE(LUOUT,28) DPBE,DPV(3,2),(DPC(7,N),N=3,NGENDP)
      WRITE(LUOUT,28) DPFG,DPG(3,2),(DPG(7,N),N=3,NGENDP)
      WRITE(LUOUT,2499)
      WRITE(LUOUT,28) (RES(J),J=4,7),RESI,RESII,RESPUL
 2900 WRITE(LUOUT,2499)
      GO TO 1000          !goto next time step K+1
 3000 CONTINUE
C
C      START OF EXHALATION; SET THE EXHALATION PARAMETERS
C
      JCYCLE=-1           !Exhalation coefficient
      FDG=1.7
      Jstatus=2           !Exhalation indicator

C   MAKE THE INHALATION P-FV CURVE AN INHALATION CURVE

       PTMMIN=PTMMIN-DPHYS
       PTMMAX=PTMMAX-DPHYS

C   SAVE THE MAXIMUM REGIONAL VOLUMES

      VMAX(1)=V(1)
      DO 3100 J=4,7
 3100 VMAX(J)=V(J)

C   CALCULATE THE END INHALATION TRACER CONCENTRATION IN EACH REGION

C      NORMALIZE THE AMOUNT TO 1.0 microcurie

       SUM=0.0
       DO 3120 J=4,7
 3120  SUM=SUM+AMOUNT(J)
       DO 3130 J=4,7
```

Program SBBOLUSD.FOR

```
 3130  AMOUNT(J)=AMOUNT(J)/SUM

       DO 3150 J=4,7
       CBOLreg(J)=AMOUNT(J)/(V(J)-RDS)
       CBOLreg(J)=1000000 * CBOLreg(J)
       WRITE(LUOUT,14) J,V(J),CBOLreg(J),AMOUNT(J)
 3150  WRITE(CRT,14) J,V(J),CBOLreg(J),AMOUNT(J)
       PAUSE'PAUSE TO LOOK AT J,V(J),CBOLreg(J),AMOUNT(J)'

       GOTO 1006
 4500  PAUSE'ITERATION NOT CONVERGING!! IGUESS=25'

 5000  CONTINUE
         STOP
       END
C-----------------------------------------------------------------------
       SUBROUTINE INTERP(Y,IMAX1,DX1,IMAX2,DX2)
       DIMENSIONY(211), X1(211), X2(601), Y2(601)
       DO 10 I1 = 1,IMAX1
            X1(I1) = (I1 -1)*DX1
   10    CONTINUE
       DO 20 I2 = 1,IMAX2
            X2(I2) = (I2-1)*DX2
   20    CONTINUE
       DO 30 I2 = 1,IMAX2
            I1 = X2(I2)/DX1 + DX1/2.0
            I1 = I1 +1
            IF(I1 .EQ. 1) I1 = 2
            IF(I1 .EQ. IMAX1) I1 = IMAX1 - 1
            U = (X2(I2) - X1(I1))/DX1
            D1 = Y(I1+1) - Y(I1-1)
            D2 = Y(I1+1) - 2.0*Y(I1) + Y(I1-1)
            Y2(I2) = Y(I1) + 0.5*D1*U + 0.5*D2*U**2
   30    CONTINUE
       DO 40 I2 = 1,IMAX2
            Y(I2) = Y2(I2)
   40    CONTINUE
       RETURN
         END
C-----------------------------------------------------------------------
       SUBROUTINE FLOWW(KMAX,KEINS,VSTART)
        COMMON FLOW(601),VOLIN(601),F(15),RTLC(15),WL(25),WD(25),
     + P1(201),V(7),AWL(7,18),H(7),Z(10),ZO(10),RCV(7),DVdt,COEFA,
     + AWD(7,18),Q(7),Pst(2,7,151),PTM(7,30),DENSITY,VISCSTY,VISK,
     + P(15),DPR(15),RES(15),DPRL(15),DMAX(25),VO(15),JCOUNT,U(7,20),
     + VOLD(15),K,ImaxFC,NUNITS,DFC,FCMIN,APGPL(25),NGEN,DD,XLMAX(24),
     + PERIOD,DT,TMAX,FCMAX,DELVOL,FDMAX(201),XKM1(7),XK(7,25),
     + DPG(7,18),RESCON(7,18),RESVIS(7,18),JINSIGN,JEXSIGN,FCFRC,TLC,
     + DPptm,ImaxFCM1,JCYCLE,PTMMIN,PTMMAX,VE,VMINLUNG,WEIGHT,DPGRAV(7),
     + FVt(201),ImaxFCP1,Jmax,TXAR(7,30),POIS(7,30),FDMAXO,SLOPEO,
     + JSTADYN,JQSTATIC,DPGRAVO(7),DFCS,DFC2,AMOUNT(7),GI,JmaxPTM,
     + NGENDP,NGENDP1,DK,QG(5),RWEIGHT(10),RVE(10),RMINVOL(10),QR(10),
     + DPV(7,30),DPC(7,30)
C
       INTEGER CRT
       CRT=0
 1     FORMAT(I3,F8.2,F8.3,F8.2)
 2     FORMAT(9F8.1)
 3     FORMAT(3X,I5)
C
C   READ THE FLOW DATA FROM THE DATA FILE CALLED FLOW.DAT
C          Jmax = # OF FLOW DATA POINTS FROM EXPERIMENTAL RECORD
C          VSTART=GAS VOLUME OF THE LUNG AT START OF BREATH
C          DTC = TIME INCREMENT FOR MEASURED FLOW DATA
C          TIDVOL = MEASURED TIDAL VOLUME
```

Program SBBOLUSD.FOR

```
C
      READ(4,1) JmaxQ,VSTART,DTC,TIDVOL
      READ(4,2) (FLOW(I),I=1,JmaxQ)
C      WRITE(CRT,1) JmaxQ,VSTART,DTC,TIDVOL
C      WRITE(CRT,2) (FLOW(I),I=1,JmaxQ)
C      PAUSE'PAUSE TO LOOK AT Jmax,VSTART,DTC,TIDVOL  AND FLOWS IN SUB FLOWW'
C
C    DETERMINE WHERE THE MEASURED DATA BECOMES NEGATIVE, JEINS.
C
      J=0
 100  J=J+1
      IF(FLOW(J).GE.0.0) GOTO 100
      JEINS=J-1
C
C    DO AN INITIAL SCALING TO MATCH THE FLOWS TO THE TIDAL VOLUME
C
      VSUM=0.0
      DO 110 J=2,JEINS
 110  VSUM=VSUM+DTC/2*(FLOW(J-1)+FLOW(J))
      SCALE = TIDVOL/VSUM
      DO 120 J=1,JmaxQ
 120  FLOW(J)=SCALE*FLOW(J)
C
C   INTERPRET FLOW DATA TO A FINER KMAX GRID.  THE TIME STEP DT FOR THIS
C   US BASED ON AN AVERAGE VOLUME INCREMENT OF DELVOL OVER INHALATION.
C
      JmaxQ1=JmaxQ-1
      TMAX=DTC*JmaxQ1
      TIMEINH = (JEINS-1)*DTC
      DT=TIMEINH*(DVdt/TIDVOL)
      KMAX1=TMAX/DT
      DT=TMAX/KMAX1
      KMAX=KMAX1+1
      DTH=DT/2.0

      CALL INTERP(FLOW,JmaxQ,DTC,KMAX,DT)

C     WRITE(CRT,2) (FLOW(I),I=1,KMAX)
C     WRITE(CRT,2) DTC,DT
C     PAUSE'PAUSE TO LOOK AT FLOWS AND DTC AND DT IN SUB FLOWW'
C
C  DETERMINE THE K WHERE THE FLOW FIRST BECOMES NEGATIVE,
C  THE END OF INHALATION (KEINS). LOOK FOR KCHECK FLOWS LESS THAN ZERO.
C
      KCHECK=3
      KC=0
      K=0
 150  K=K+1
C      WRITE(CRT,3)K
      IF(FLOW(K).GT.0) GOTO 200
      KC=KC+1
      IF(KC.EQ.KCHECK) GOTO 250
      GOTO 150
 200  KC=0
      GOTO 150
 250  KEINS=K-(KCHECK-1)
C      PAUSE'PAUSE TO LOOK AT ALL THE Ks'
C
C  INTEGRATE FLOW TO GET TIDAL VOLUME BASED ON FLOW DATA (FTIDVOL)
C
      KEINS1=KEINS-1
      FTIDVOL=0.0
      DO 300 K=1,KEINS1
 300  FTIDVOL=FTIDVOL+DTH*(FLOW(K)+FLOW(K+1))
```

Program SBBOLUSD.FOR

```
C
C   NORMALIZE FLOWS WITH RESPECT TO THE MEASURED TIDAL VOLUME
C
      SCALE=TIDVOL/FTIDVOL
      DO 400 K=1,KMAX
 400  FLOW(K)=SCALE*FLOW(K)
C
C  CALCULATE THE INHALED VOLUME (VOLIN) BASED ON THE SCALED FLOWS
C
      VOLIN(1)=0.0
      DO 500 K=2,KMAX
 500  VOLIN(K)=VOLIN(K-1)+DTH*(FLOW(K)+FLOW(K+1))



C      WRITE(CRT,2) (FLOW(K),K=1,KMAX)
C      WRITE(CRT,2) (VOLIN(K),K=1,KMAX)
C      PAUSE'PAUSE TO LOOK AT VOLIN(K) IN SUB FLOWW'

      RETURN
      END
C-----------------------------------------------------------------------
      SUBROUTINE TRANMPR(OMEGA,DPFG)
       COMMON FLOW(601),VOLIN(601),F(15),RTLC(15),WL(25),WD(25),
     + P1(201),V(7),AWL(7,18),H(7),Z(10),Z0(10),RCV(7),DVdt,COEFA,
     + AWD(7,18),Q(7),Pst(2,7,151),PTM(7,30),DENSITY,VISCSTY,VISK,
     + P(15),DPR(15),RES(15),CPRL(15),DMAX(25),V0(15),JCOUNT,U(7,20),
     + VOLD(15),K,ImaxFC,NUNITS,DFC,FCMIN,APGPL(25),NGEN,DD,XLMAX(24),
     + PERIOD,DT,TMAX,FCMAX,DELVOL,FDMAX(201),XKM1(7),XK(7,25),
     + DPG(7,18),RESCON(7,18),RESVIS(7,18),JINSIGN,JEXSIGN,FCFRC,TLC,
     + DPptm,ImaxFCM1,JCYCLE,PTMMIN,PTMMAX,VE,VMINLUNG,WEIGHT,DPGRAV(7),
     + FVt(201),ImaxFCP1,Jmax,TXAR(7,30),POIS(7,30),FDMAX0,SLOPE0,
     + JSTADYN,JQSTATIC,DPGRAV0(7),DFCS,DFC2,AMOUNT(7),GI,JmaxPTM,
     + NGENDP,NGENDP1,DK,QG(5),RWEIGHT(10),RVE(10),RMINVOL(10),QR(10),
     + DPV(7,30),DPC(7,30)
C
       DIMENSION SUMDPG(10),II(10)
      INTEGER CRT
      CRT=0
  1   FORMAT(1X,9E8.2)
  2   FORMAT(3X,2I4,F8.2)
C        PAUSE'PAUSE AFTER YOU ENTER TRANSMP'
      IF(K.NE.100) GOTO 200
      KKK=300
 200  CONTINUE
      JmaxM1 = Jmax-1
      IF (K.EQ.1) GO TO 4
      IF (JQSTATIC.EQ.1) GO TO 8
      IF (JSTADYN.EQ.1) GO TO 8
C
C  FOR THE STATIC CASE PTM = PARENCHYMAL RECOIL  PRESSURE
C
  4   CONTINUE
      DO 6 J=4,7

      DO 5 N=1,NGENDP
  5   PTM(J,N)=P(J)
C        WRITE(CRT,1) (PTM(J,N),N=1,NGENDP)
C        PAUSE'PAUSE TO LOOK AT PTMs'
  6   CONTINUE
      GOTO 100
  8   CONTINUE
C        WRITE(CRT,2)JCYCLE
C        PAUSE'PAUSE TO LOOK AT JCYCLE'
      IF(JCYCLE.LT.0) GO TO 10
```

Program SBBOLUSD.FOR

```
C
C                     DYNAMIC CASE
C
C            SET XKM=1 AND XK=1 DURING INHALATION
C
      DO 9 J=4,7
      XKM1(J)=1.00
      DO 9 N=1,NGENDP
    9 XK(J,N)=1.0
      XK(2,2)=1.0
      XK(3,2)=1.0
      GO TO 75
   10 CONTINUE
C
C     CALCULATION OF XKM1=KM-1. Km IS A FUNCTION OF THE LOCAL
C     COMPLIANCE OF THE PARENCHYMA. SEE EQ.3.16.
C
      DO 50 J=4,7
      RFV=V(J)/RTLC(J)
      JI=II(J)
      FR=RFV-FVt(JI)
      IF(FR)31,33,32
   31 JI=JI-1
      IF(FVt(JI).LE.RFV) GO TO 33
      GO TO 31
   32 IF(JI.GE.JmaxM1) GO TO 33
      JI=JI+1
      IF(FVt(JI).GE.RFV) GO TO 33
      GO TO 32
   33 II(J) = JI
      IF(JI.GE.Jmax) JI=JmaxM1
      C=(FVt(JI+1)-FVt(JI))/DPptm
      XKM1(J)=2.0/(1.0+OMEGA*C)
C
C    CALCULATION OF XK - THE MEASURE OF THE INTERDEPENDENCE
C    BETWEEN THE LOCAL PARENCHYMA AND THE BRONCHIAL WALL.
C
C                     GENS 3-NGENDP
C
      DO 40 N=3,NGENDP
      AWDUE=DMAX(N)*RFV**(1./3.)
      DSTAR = AWD(J,N)/AWDUE
      IF(DSTAR.GT.1.0) DSTAR = 1.0
      IF(DSTAR.LE..7) DSTAR = .7
   40 XK(J,N)=XKM1(J)*(1.25/(1.0+((DSTAR-.7)/.15)**2)-.25)+1.0

C        WRITE(CRT,1)         (XK(JJ,N),N=3,NGENDP)
C        PAUSE'PAUSE TO LOOK AT (XK(JJ,N),N=3,NGENDP)'
 50    CONTINUE
C
C                              GENERATION 2
C

      DO 60 J=2,3
      IF(J.EQ.2) JJ=5
      IF(J.EQ.3) JJ=6
      RFV=V(JJ)/RTLC(JJ)
      AWDUE=DMAX(N)*RFV**(1./3.)
      DSTAR = AWD(J,N)/AWDUE
      IF(DSTAR.GE.1) DSTAR = 1.0
      IF(DSTAR.LE..7) DSTAR = .7
    . XK(J,2)=XKM1(JJ)*(1.25/(1.0+((DSTAR-.7)/.15)**2)-.25)+1.0
C        WRITE(CRT,2) XK(J,2)
   60 CONTINUE
C        PAUSE'PAUSE TO LOOK AT THE XKs'
```

```
  75  CONTINUE
C
C     CALCULATE THE TRANSMURAL PRESSURE ACTING ON EACH
C          BRONCHIAL WALL. SEE EQ 3.15
C
C               GENERATIONS 3-NGENDP
C
      N1=NGENDP-2
      DO 85 J=4,7
      SUMDPG(J)=0.0
      DO 80 N=NGENDP,3,-1
      SUMDPG(J)=SUMDPG(J)-DPG(J,N)
  80  PTM(J,N)=P(J)+JCYCLE/XK(J,N)*SUMDPG(J)
  85  CONTINUE
C
C   CALCULATE PTM IN GENS 1 AND 2 AND THE TRACHEA
C
      SUMDPG(5)=SUMDPG(5)-DPG(2,2)
      SUMDPG(6)=SUMDPG(6)-DPG(3,2)
      PTM(2,2)=P(5)+JCYCLE/XK(2,2)*SUMDPG(5)
      PTM(3,2)=P(6)+JCYCLE/XK(3,2)*SUMDPG(6)
      SUMDPG(5)=SUMDPG(5)-DPFG
      PTM(1,1)=P(5)+JCYCLE*SUMDPG(5)
      PTMTRACH=P(5)+JCYCLE*SUMDPG(5)
  100 CONTINUE
      RETURN
      END
C-----------------------------------------------------------------
      SUBROUTINE STATVOL(FClung,Jstatus)
C
C   THIS SUBROUTINE CALCULATES THE FOUR REGIONAL VOLUMES USING AN
C   ITERATION SCHEME (METHOD OF FALSE POSITION).  THE PROCEDURE
C   STARTS WITH A GUESS OF P(4), THE PRESSURE AT THE CG OF THE
C   HIGHEST REGION.  THE OTHER PRESSURES ARE CALCULATED USING DPRAVO
C   AND THE RELATIVE VOLUME EXPANSION OF THE ENTIRE LUNG.  THE REGIONAL
C   AND TOTAL VOLUMES ARE THEN CALCULATED. THE CALCULATED VOLUME IS THE
C   COMPARED TO THE REQUIRED VOLUME,WHICH IS THE ERROR. THE ITERATION
C   SCHEME PROCEEDS ONCE TWO GUESSES HAVE BEEN MADE.
C
       COMMON FLOW(601),VOLIN(601),F(15),RTLC(15),WL(25),WD(25),
     + P1(201),V(7),AWL(7,18),H(7),Z(10),ZO(10),RCV(7),DVdt,COEFA,
     + AWD(7,18),Q(7),Pst(2,7,151),PTM(7,30),DENSITY,VISCSTY,VISK,
     + P(15),DPR(15),RES(15),DPRL(15),DMAX(25),VO(15),JCOUNT,U(7,20),
     + VOLD(15),K,ImaxFC,NUNITS,DFC,FCMIN,APGPL(25),NGEN,DD,XLMAX(24),
     + PERIOD,DT,TMAX,FCMAX,DELVOL,FDMAX(201),XKM1(7),XK(7,25),
     + DPG(7,18),RESCON(7,18),RESVIS(7,18),JINSIGN,JEXSIGN,FCFRC,TLC,
     + DPptm,ImaxFCM1,JCYCLE,PTMMIN,PTMMAX,VE,VMINLUNG,WEIGHT,DPGRAV(7),
     + FVt(201),ImaxFCP1,Jmax,TXAR(7,30),POIS(7,30),FDMAXO,SLOPEO,
     + JSTADYN,JQSTATIC,DPGRAVO(7),DFCS,DFC2,AMOUNT(7),GI,JmaxPTM,
     + NGENDP,NGENDP1,DK,QG(5),RWEIGHT(10),RVE(10),RMINVOL(10),QR(10),
     + DPV(7,30),DPC(7,30)
C
      DIMENSION FCR(7),VR(7),PR(7)
      INTEGER R,CRT
      CRT=0
  1   FORMAT(3X,I5,4F10.2)
  2   FORMAT(2X,I3,8E9.3)
  3   FORMAT(2X,8E10.3)
C
      IGUESS=1
      CHECKV=2.0     !The calculated volume must be within CHECKV cc's
C    .          of the actual volume.
C
C USE P-FVA (PRESS-FRACTIONAL VOLUME OF THE ACINUS) TO GET THE 1st GUESS
C           FOR P(4) (PRESSURE AT THE CG OF REGION 4), P1
```

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C
      J=Jmax
100   J=J-1
      IF(FVt(J).GT.FCLUNG) GOTO 100
      PG1=PTMMIN+(J-1)*DPptm
      PGRAV=DPGRAV(4)+DPGRAV(5)+DPGRAV(6)
      PR(4)= PG1+PGRAV/2         !PG1 IS THE FIRST GUESS
C      WRITE(CRT,2) J,FVt(J),FCLUNG,PTMMIN,PTMMAX,DPptm,PR(4)
C      PAUSE'PAUSE TO LOOK AT J,FVt(J),FCLUNG,PTMMIN,DPptm,PR(4)'
C
C  CALCULATE THE PRESSURES IN THE THREE LOWER REGIONS
C
      DO 200 R=5,7
 200  PR(R) = PR(R-1)-DPGRAV(R-1)
C      WRITE(CRT,3) (PR(R),R=4,7)
C      PAUSE'PAUSE TO LOOK AT PR(R),R=4,7 1st GUESS'
C
C  CALCULATE THE LUNG VOLUME FOR THESE PRESSURES
C
      VOLCALC=0

      DO 300 R=4,7


      I=1
250   I=I+1
      IF (PR(R).GT.Pst(Jstatus,R,I)) GOTO 250

      UU=(PR(R)-Pst(Jstatus,R,I-1))
     *          /(Pst(Jstatus,R,I)-Pst(Jstatus,R,I-1))

 280  FCR(R)= FCMIN + (I-2)*DFC +UU*DFC



      VR(R)=FCR(R)*RTLC(R)
      VOLCALC = VOLCALC+VR(R)   !VOLCALC is calculated lung volume

C      WRITE(CRT,2) I,Pst(Jstatus,R,I),PR(R),FCR(R),RTLC(R),FCMIN,DFC,U,VR(R)
C      PAUSE'I,PST,PR(R),FCR(R),RTLC(R),FCMIN,DFC,U,VR(R)- 1st guess'
 300  CONTINUE
C      WRITE(CRT,3) V(1),VOLCALC,ERR1
C      PAUSE' V(1),VOLCALC,ERR1 AFTER 1ST GUESS'
C
C  OBTAIN THE NEXT GUESS FOR P(4), PG2
C
      ERR1=VOLCALC-V(1)       !ERR1 is the error for the 1st guess, P1
      IF(ERR1)400,500,450
 400  PG2=PG1+.25    !VOLCALC too small, make PG2 bigger than PG1
      GO TO 500
 450  PG2=PG1-.25    !VOLCALC too big, make PG2 less than PG1
C
C  CALCULATE THE PRESSURES IN THE THREE LOWER REGIONS
C
 500  PR(4)=PG2
      DO 550 R=5,7
 550  PR(R)=PR(R-1)-DPGRAV(R-1)
C      WRITE(CRT,3) (PR(R),R=4,7)
C      PAUSE'PAUSE TO LOOK AT PR(R),R=4,7 FOR NEXT GUESS'
C
C  CALCULATE THE REGIONAL AND TOTAL LUNG VOLUMES FOR THESE PRESSURES
C
      VOLCALC=0
      DO 700 R=4,7
      I=1
```

```
 650  I=I+1
      IF (PR(R).GT.Pst(Jstatus,R,I)) GOTO 650

      UU=(PR(R)-Pst(Jstatus,R,I-1))
     *        /(Pst(Jstatus,R,I)-Pst(Jstatus,R,I-1))

 690  FCR(R)= FCMIN + (I-2)*DFC + UU*DFC
      VR(R)=FCR(R)*RTLC(R)
      VOLCALC = VOLCALC+VR(R)
C     WRITE(CRT,2) I,Pst(Jstatus,R,I),PR(R),FCR(R),RTLC(R),FCMIN,DFC,UU,VR(R)
C     PAUSE'PAUSE TO SEE I,PST,PR(R),FC(R),RTLC(R),FCMIN,DFC,UU,VR(R)'
 700  CONTINUE
      ERR2=VOLCALC-V(1)
      AERR2=ABS(ERR2)
      IF(AERR2.LT.CHECKV) GOTO 1000
C     WRITE(CRT,1) IGUESS,PG1,PG2,VOLCALC,ERR2
C     PAUSE'IGUESS,PG1,PG2,VOLCALC,ERR2'
C
C  USE THE ITERATION SCHEME TO MAKE THE NEXT GUESS, PG3
C
 750  CONTINUE
      PG3=(PG1*ERR2-PG2*ERR1)/(ERR2-ERR1)
      PG1=PG2
      PG2=PG3
      ERR1=ERR2
      IGUESS=IGUESS+1
      IF(IGUESS.LT.25) GOTO 800
      PAUSE'IGUESS=25 IN STATVOL, TYPE CONTROL C AND GET OUT!!!'
      KKK=1
 800  GOTO 500
 1000  CONTINUE

       DO 1100 R=4,7
       P(R)=PR(R)
       RVE(R)=(VR(R)+RWEIGHT(R))/RMINVOL(R)
 1100  V(R)=VR(R)
       V(2)=V(4)+V(5)
       V(3)=V(6)+V(7)
       RETURN
       END
C-----------------------------------------------------------------
      SUBROUTINE AIRWAY(TRCDMAX,TRCDIAM)
C
C THIS SUBROUTINE CALCULATES THE AIRWAY LENGTHS AND DIAMETERS DURING THE
C BREATH.  LENGTHS ARE BASED ON THE REGIONAL EXPANSION AND DIAMETERS ARE
C  BASED ON THE TRANSMURAL PRESSURE THAT ACTS ON THE AIRWAY.
C
       COMMON FLOW(601),VOLIN(601),F(15),RTLC(15),WL(25),WD(25),
     + P1(201),V(7),AWL(7,18),H(7),Z(10),Z0(10),RCV(7),DVdt,COEFA,
     + AWD(7,18),Q(7),Pst(2,7,151),PTM(7,30),DENSITY,VISCSTY,VISK,
     + P(15),DPR(15),RES(15),DPRL(15),DMAX(25),V0(15),JCOUNT,U(7,20),
     + VOLD(15),K,ImaxFC,NUNITS,DFC,FCMIN,APGPL(25),NGEN,DD,XLMAX(24),
     + PERIOD,DT,TMAX,FCMAX,DELVOL,FDMAX(201),XKM1(7),XK(7,25),
     + DPG(7,18),RESCON(7,18),RESVIS(7,18),JINSIGN,JEXSIGN,FCFRC,TLC,
     + DPptm,ImaxFCM1,JCYCLE,PTMMIN,PTMMAX,VE,VMINLUNG,WEIGHT,DPGRAV(7),
     + FVt(201),ImaxFCP1,Jmax,TXAR(7,30),POIS(7,30),FDMAX0,SLOPE0,
     + JSTADYN,JQSTATIC,DPGRAV0(7),DFCS,DFC2,AMOUNT(7),GI,JmaxPTM,
     + NGENDP,NGENDP1,DK,QG(5),RWEIGHT(10),RVE(10),RMINVOL(10),QR(10),
     + DPV(7,30),DPC(7,30)

       INTEGER CRT,R
       CRT=0
  1    FORMAT(2X,8E9.3)
  2    FORMAT(2X,I5)
  3    FORMAT(2X,I4,4F9.3)
```

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 4    FORMAT(2X,2I5,F10.4)
C
C    CALCULATE THE LENGTHS FOR GENS 1 AND 2
C
C     WRITE(CRT,2) R
      AWL(1,1) = (V(1)/RTLC(1))**(1./3.) * XLMAX(1)
      AWL(2,2) = (V(2)/RTLC(2))**(1./3.) * XLMAX(2)
      AWL(3,2) = (V(3)/RTLC(3))**(1./3.) * XLMAX(2)
C     PAUSE'IN AIRWAY, AFTER AWL(3,2)'
C
C    CALCULATE THE LENGTHS FOR GENS 3-23
C
      DO 100 R=4,7
      SCALE=(V(R)/RTLC(R))**(1./3.)
C     WRITE(CRT,3) R,V(R),RTLC(R),SCALE
      DO 50 N=3,NGENDP
 50   AWL(R,N)=SCALE*XLMAX(N)
C     WRITE(CRT,2) R
C     WRITE(CRT,1) (AWL(R,N),N=3,NGENDP)
C     PAUSE'PAUSE TO LOOK AT THE AWLS'
 100  CONTINUE
C
C    CALCULATE THE AIRWAY DIAMETERS FOR THE TRACHEA AND GEN 1 (AREA 2)
C
      TRCDIAM= 0.80+0.056*PTM(1,1) * TRCDMAX
      AWD(1,1)=0.82+0.050*PTM(1,1) * DMAX(1)
C
C    CALCULATE THE AIRWAY DIAMETERS FOR GEN's 2-NGENDP
C
      PTM(4,2)=PTM(2,2)   !calculate PTM(j,2) for use in the do loop
      PTM(5,2)=PTM(2,2)
      PTM(6,2)=PTM(3,2)
      PTM(7,2)=PTM(3,2)
      if(k.LT.237) GOTO 119
      KKK=1
 119  CONTINUE

      DO 200 R=4,7
      DO 160 N=2,NGENDP
      IF(PTM(R,N).LT.0.0) GOTO 150

C   CALCULATION OF AIRWAY DIAMETERS FOR POSITIVE PTM's
      I=1
 120  I=I+1
      PTT=PTMMIN+(I-1)*DPptm
      IF(I.EQ.JmaxPTM) GOTO 130
      IF(PTM(R,N).GT.PTT) GOTO 120
      UU=(PTM(R,N)-(PTT-DPptm))/DPptm
      SCALE=FDMAX(I-1)+UU*(FDMAX(I)-FDMAX(I-1))
      GOTO 140
 130  UU=(PTM(R,N)-(PTT-DPptm))/DPptm
      SCALE= FDMAX(JmaxPTM)+UU*(FDMAX(JmaxPTM)-FDMAX(JmaxPTM-1))
 140  AWD(R,N)=SCALE*DMAX(N)
      GOTO 160

C   CALCULATION OF AIRWAY DIAMETERS FOR NEGATIVE PTM'S
 150  AWD(R,N)=FDMAX0*DMAX(N)/(1.-PTM(R,N)/SLOPE0)**(1./3.)

 160  CONTINUE
      IF(K.LT.1000) GOTO 200
      WRITE(CRT,2) R
      WRITE(CRT,1) (AWD(R,N),N=2,NGENDP)
      WRITE(CRT,1) (PTM(R,N),N=2,NGENDP)
      PAUSE'PAUSE TO LOOK AT THE AWDS'
 200  CONTINUE
```

```
      AWD(2,2)=AWD(5,2)
      AWD(3,2)=AWD(6,2)
      RETURN
      END
C-----------------------------------------------------------------
      SUBROUTINE PRESSLS(A1,A2,LSIT,QLATEST)
       COMMON FLOW(601),VOLIN(601),F(15),RTLC(15),WL(25),WD(25),
     + P1(201),V(7),AWL(7,18),H(7),Z(10),Z0(10),RCV(7),DVdt,COEFA,
     + AWD(7,18),Q(7),Pst(2,7,151),PTM(7,30),DENSITY,VISCSTY,VISK,
     + P(15),DPR(15),RES(15),DPRL(15),DMAX(25),V0(15),JCOUNT,U(7,20),
     + VOLD(15),K,ImaxFC,NUNITS,DFC,FCMIN,APGPL(25),NGEN,DD,XLMAX(24),
     + PERIOD,DT,TMAX,FCMAX,DELVOL,FDMAX(201),XKM1(7),XK(7,25),
     + DPG(7,18),RESCON(7,18),RESVIS(7,18),JINSIGN,JEXSIGN,FCFRC,TLC,
     + DPptm,ImaxFCM1,JCYCLE,PTMMIN,PTMMAX,VE,VMINLUNG,WEIGHT,DPGRAV(7),
     + FVt(201),ImaxFCP1,Jmax,TXAR(7,30),POIS(7,30),FDMAX0,SLOPE0,
     + JSTADYN,JQSTATIC,DPGRAV0(7),DFCS,DFC2,AMOUNT(7),GI,JmaxPTM,
     + NGENDP,NGENDP1,DK,QG(5),RWEIGHT(10),RVE(10),RMINVOL(10),QR(10),
     + DPV(7,30),DPC(7,30)
C
C       LSIT=1 CALCULATE TOTAL AIRWAY AREAS AND PRESSURES FOR ALL
C            AIRWAYS FROM NGENDP TO GENERATION 2 (1st CALL FROM MAIN)
C       LSIT=2 CALCULATE PRESSURE CHANGES IN ALL AIRWAYS (LATER CALLS
C            FROM PROGRAM MAIN
C       LSIT=3 CALCULATE PRESSURE CHANGES IN AREAS A1 AND A2 WHEN
C            CALLED FROM REGFLOW
C

      INTEGER CRT,AREA,A1,A2
      CRT=0
    1 FORMAT(2X,7E10.3)
    2 FORMAT(5X,3I10)
C      WRITE(CRT,2)LSIT,NGENDP,NGENDP1
C      PAUSE'PAUSE AFTER YOU ENTER PRESSLS AND SHOW LSIT,NGENDP,NGENDP1'
C
      PIE=3.1416
      PIE4=PIE/4
      SQ2PI= 1.4142*PIE*VISCSTY
      NGENDP1=NGENDP+1               !calculates velocity for an extra gen
      IF(LSIT.NE.1)GOTO 210     !areas already calculated
C
C     CALCULATE THE CROSS-SECTIONAL AREA OF THE AIRWAYS IN EACH
C          GENERATION OF EACH OF THE AREAS
C
C        CROSS-SECTIONAL AREA OF GENERATIONS 3-NGENDP
C
      DO 150 AREA=A1,A2
      DO 100 N=3,NGENDP
      NAWPG=(2**N)/4    !# of Airways Per Generation in areas 4,5,6 and 7
      TXAR(AREA,N)=NAWPG*PIE4*AWD(AREA,N)**2
  100 CONTINUE
      IF(K.LT.1000) GOTO 150
      WRITE(CRT,2) AREA
      WRITE(CRT,1) (TXAR(AREA,J),J=3,NGENDP1)
      PAUSE'PAUSE TO LOOK AT THE AREAs OF GENs 3-21'
  150 CONTINUE
C
C     AREA OF GENERATION 2
C
       NAWPG=2              !Number of Airways in gen 2 in AREAs 2 and 3
       N=2
      DO 200 AREA=2,3
     ·TXAR(AREA,N)=NAWPG*PIE4*(AWD(AREA,N))**2
C     WRITE(CRT,1) TXAR(AREA,2)
C     PAUSE'PAUSE TO LOOK AT THE AREA IN SECOND GENERATION'
  200 CONTINUE
```

```
 210  CONTINUE
C
C       CALCULATE THE GAS VELOCITIES IN THE AIRWAYS
C   NOTE: THE GAS VELOCITIES IN THE 1st GEN AND THE TRACHEA ARE NOT
C       REQUIRED FOR THE ITERATION.  THESE ARE CALCULATED IN THE
C       MAIN PROGRAM TO LOOK AT OVERALL PRESSURES
C
C  FIRST CALCULATE THE FLOWS TO ALL AREAS BASED ON THIS GUESS-THE
C  VARIABLE USED IS QR.  IT IS USED ONLY IN PRESSLS AND REGVOL TO
C  SAVE THE FLOWS BASED ON THE LATEST GUESS FOR Q(2) FROM PROG MAIN
C
      IF(LSIT.EQ.3) GOTO 220
      QR(2)=QLATEST
      QR(3)=Q(1)-QR(2)
      QR(4)=Z(4)*2*QR(2)
      QR(5)=QLATEST-QR(4)
      QR(6)=Z(6)*2*QR(3)
      QR(7)=QR(3)-QR(6)
      GOTO 240
 220  CONTINUE
      QR(A1)=QLATEST          !need flows for A1 and A2 when called from REGFLOW
      J=A2/2
      QR(A2)=QR(J)-QR(A1)
 240  CONTINUE
C      WRITE(CRT,1) (QR(AREA),AREA=A1,A2)
C      PAUSE'PAUSE TO LOOK AT THE REGIONAL FLOWS IN PRESSLS'
C
C         CALCULATE THE GAS VELOCITIES IN GEN 3-NGENDP
C
      DO 275 AREA=A1,A2
      DO 250 N=3,NGENDP
      U(AREA,N)=QR(AREA)/TXAR(AREA,N)
 250  CONTINUE
C     WRITE(CRT,2) AREA
C     WRITE(CRT,1) (U(AREA,J),J=3,NGENDP1)
C     PAUSE'PAUSE TO LOOK AT THE VELOCITIES IN GENs 3-21'
 275  CONTINUE
      IF(LSIT.EQ.3) GOTO 310
C
C     GAS VELOCITIES IN GEN 2 IN AREAS 2 AND 3
C
      N=2
      DO 300 AREA=2,3
      U(AREA,N)=QR(AREA)/TXAR(AREA,N)
C     WRITE(CRT,2) AREA
C     WRITE(CRT,1) QR(AREA),U(AREA,2)
C     PAUSE'PAUSE TO LOOK AT QR AND U FOR THE SECOND GENERATION'
 300  CONTINUE
C
C      GAS VELOCITY IN GENERATION 1
C
      U(1,1)=ABS(Q(1)/(1.571*AWD(1,1)**2))
C
 310  CONTINUE
C
C   CALCULATE THE VISCOUS LOSSES IN THE FLOW USING A CORRECTED EQ 3.26
C     DPV IS NEGATIVE FOR INHALATION;POSITIVE FOR EXHALATION
C     INHALATION FLOWS PRODUCE POSITIVE VELOCITIES
C     EXHALATION FLOWS PRODUCE NEGATIVE VELOCITIES
C
C           VISCOUS LOSSES IN GENERATIONS 3-NGDP
C
      DO 400 AREA=A1,A2
      DO 350 N=3,NGENDP
      ABVEL=ABS(U(AREA,N))
```

```
      RE=ABVEL*AWD(AREA,N)/VISK
      W1=(32.0*AWL(AREA,N))/(RE*AWD(AREA,N))
      W2=DENSITY*U(AREA,N)**2
      POIS(AREA,N)=GI*W1*W2
      W3=((RE*AWD(AREA,N)/AWL(AREA,N))**(1./2.))/3.2703
      POIS(AREA,N)=W3*POIS(AREA,N)
      SQRE=(RE*AWD(AREA,N)/AWL(AREA,N))**(1./2.)
      COEFB=-U(AREA,N)*AWL(AREA,N)/AWD(AREA,N)**2
      DPV(AREA,N)=COEFA*COEFB*SQRE
 350  DPV(AREA,N)=GI*DPV(AREA,N)   !convert DP from dynes/cm**2 to cmH2O
C     WRITE(CRT,2) AREA
C     WRITE(CRT,1) (DPV(AREA,N),N=3,NGENDP)
C     PAUSE'PAUSE TO LOOK AT THE DPVs IN GENs 3-NGENDP'
 400  CONTINUE
      IF(LSIT.EQ.3) GOTO 510
C
C       VISCOUS LOSSES IN GENERATION 2
C
      N=2
      DO 500 AREA=2,3
      ABVEL=ABS(U(AREA,N))
      RE=ABVEL*AWD(AREA,N)/VISK
      SQRE=(RE*AWD(AREA,N)/AWL(AREA,N))**(1./2.)
      COEFB=-U(AREA,N)*AWL(AREA,N)/AWD(AREA,N)**2
      DPV(AREA,N)=COEFA*COEFB*SQRE
      DPV(AREA,N)=GI*DPV(AREA,N) !convert DP from dynes/cm**2 to cmH2O
C     WRITE(CRT,2) AREA
C     WRITE(CRT,1) RE,DPV(AREA,N)
 500  CONTINUE
C     PAUSE'PAUSE TO LOOK AT RE AND DPVs IN GEN 2'
 510  CONTINUE
C
C  CALCULATE THE CONVECTIVE ACCELERATIONS USING EQ 3.28
C     DPC IS POSITIVE FOR INHALATION;POSITIVE FOR EXHALATION
C
      DO 600 AREA=A1,A2
      DO 550 N=3,NGENDP
      DPC(AREA,N)=DK*(U(AREA,N-1)**2-U(AREA,N)**2)
 550  DPC(AREA,N)=GI*DPC(AREA,N)!convert dynes/cm**2 into cmH2o
C     WRITE(CRT,2) AREA
C     WRITE(CRT,1) (DPC(AREA,N),N=3,NGENDP)
C     PAUSE'PAUSE TO LOOK AT THE DPC FOR GENs 3-NGENDP'
 600  CONTINUE
      IF(LSIT.EQ.3) GOTO 710
C
C          CONVECTIVE ACCELERATIONS IN GENERATION 2
C
      N=2
      DO 700 AREA=2,3
      DPC(AREA,N)=DK*(U(AREA,N-1)**2-U(AREA,N)**2)
      DPC(AREA,N)=GI*DPC(AREA,N)!convert dynes/cm**2 into cmH2o
C     WRITE(CRT,2) AREA
C     WRITE(CRT,1) DPC(AREA,N)
 700  CONTINUE
C     PAUSE'PAUSE TO LOOK AT DPC FOR GEN 2'
 710  CONTINUE
C
C   CALCULATE THE TOTAL PRESSURE CHANGE IN EACH REGION (DPRL)
C           AND IN GENERATION 2 (DPG)
C
C                   CALCULATION OF DPR
C
      DO 800 AREA=A1,A2
      DPR(AREA)=0.0
      DO 750 N=3,NGENDP
```

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```
      DPG(AREA,N)=DPV(AREA,N)+DPC(AREA,N)  !DPG=DP in a gen
 750  DPR(AREA)=DPR(AREA)+DPG(AREA,N)
 800  CONTINUE
C      WRITE(CRT,1) (DPR(AREA),AREA=A1,A2)
C      PAUSE'PAUSE TO LOOK AT THE DPR CALCULATED IN PRESSLS'
      IF(LSIT.EQ.3) GOTO 910
C
C           CALCULATION OF DPG FOR GENERATION 2
C
      N=2
      DO 900 AREA=2,3
      DPG(AREA,2)=DPV(AREA,N)+DPC(AREA,N)
C      WRITE(CRT,2) AREA
C      WRITE(CRT,1) DPG(AREA,N)
 900   CONTINUE
C      PAUSE' PAUSE TO LOOK AT DPGs FOR AREAS 2 AND 3 CALC IN PRESSLS'
 910  CONTINUE
      RETURN
      END
C-----------------------------------------------------------------------
      SUBROUTINE  REGFLOW(CHECKP)
       COMMON FLOW(601),VOLIN(601),F(15),RTLC(15),WL(25),WD(25),
     + P1(201),V(7),AWL(7,18),H(7),Z(10),ZO(10),RCV(7),DVdt,COEFA,
     + AWD(7,18),Q(7),Pst(2,7,151),PTM(7,30),DENSITY,VISCSTY,VISK,
     + P(15),DPR(15),RES(15),DPRL(15),DMAX(25),VO(15),JCOUNT,U(7,20),
     + VOLD(15),K,ImaxFC,NUNITS,DFC,FCMIN,APGPL(25),NGEN,DD,XLMAX(24),
     + PERIOD,DT,TMAX,FCMAX,DELVOL,FDMAX(201),XKM1(7),XK(7,25),
     + DPG(7,18),RESCON(7,18),RESVIS(7,18),JINSIGN,JEXSIGN,FCFRC,TLC,
     + DPptm,ImaxFCM1,JCYCLE,PTMMIN,PTMMAX,VE,VMINLUNG,WEIGHT,DPGRAV(7),
     + FVt(201),ImaxFCP1,Jmax,TXAR(7,30),POIS(7,30),FDMAXO,SLOPEO,
     + JSTADYN,JQSTATIC,DPGRAVO(7),DFCS,DFC2,AMOUNT(7),GI,JmaxPTM,
     + NGENDP,NGENDP1,DK,QG(5),RWEIGHT(10),RVE(10),RMINVOL(10),QR(10),
     + DPV(7,30),DPC(7,30)
C
C  THIS SUBROUTINE CALCULATES THE FLOWS WITHIN AREAS 2 AND 3 FOR A GIVEN
C  Q(2) AND Q(3) THAT HAVE BEEN DEFINED IN THE MAIN PROGRAM
C
C
      INTEGER AREA,A1,A2,CRT
  1   FORMAT(2X,7E10.3)
  2   FORMAT(5X,3I5)

      CRT=0
      LSIT=3
C     PAUSE'PAUSE AFTER YOU ENTER REGFLOW'
C     WRITE(CRT,1) P(4),P(5),P(6),P(7)
C     PAUSE'PAUSE AFTER YOU PRINT YOUR Ps'
C
      DO 1000 AREA=2,3
      A1=2*AREA                        !A1 is AREA 4 or 6
      A2=A1+1                          !A2 is AREA 5 or 7
C      WRITE(CRT,2) A1,A2,AREA
C      PAUSE'PAUSE TO LOOK AT A1,A2,AREA AS YOU ENTER REGFLOW'
C
C  THE ITERATION DETERMINES THE FLOW IN AREA A1.  NEED TWO GUESSES FOR
C  THE FLOW IN AREA A1 TO GET THE ITERATION PROCESS (METHOD OF FALSE
C  POSITION) STARTED.  FIRST GUESS IS Q2 AND IT IS BASED ON THE LATEST
C  GUESS FOR Q(2) FROM PROG MAIN.  QRs ARE CALCULATED IN PRESSLS FOR
C  THE CASE LSIT=1.
C
      Q2=QR(A1)
C
C  NOTE:THE DPRs FOR THIS GUESS WERE CALCULATED JUST BEFORE THE MAIN
C  PROGRAM ENTERED REGFLOW.  THE ITERATION FUNCTION (F)IS THE ERROR
C  WHEN EQ. 3.32 IS NOT PROPERLY BALANCED.
```

Program SBBOLUSD.FOR

```
C
      G=P(A1)-P(A2)-DPGRAV(A1)
      F2=G+(DPR(A2)-DPR(A1))
      AF2=ABS(F2)
      IF(AF2.LT.CHECKP)GOTO 500
C
C    GET SECOND GUESS FOR FLOW TO AREA A1
C
      IF(F2.GT.0.00) Q3=1.02*Q2
      IF(F2.LT.0.00) Q3=0.98*Q2
C      WRITE(CRT,1) Q2,Q3,F2
C      PAUSE'PAUSE TO LOOK AT Q2,Q3,F2 FROM THE FIRST GUESS FOR Q(A1)'
      LCOUNT=1
 200  F1=F2
      Q1=Q2
      Q2=Q3
      CALL PRESSLS(A1,A2,LSIT,Q2)
      F2=G+(DPR(A2)-DPR(A1))
      AF2=ABS(F2)
      LCOUNT=LCOUNT+1
C      WRITE(CRT,2) LCOUNT
C      WRITE(CRT,1) Q1,Q2,F1,F2
C      PAUSE'PAUSE AFTER Q1,Q2,F1,F2'
      IF(AF2.LT.CHECKP) GOTO 500
C
C    GET NEXT VALUE OF FLOW TO AREA A1
C
      Q3=(Q1*F2-Q2*F1)/(F2-F1)
      GOTO 200
  500 CONTINUE
      Q(A1)=Q2
      Q(A2)=QG(AREA)-Q2
C      WRITE(CRT,2) A1,A2
C      WRITE(CRT,1) Q(A1),Q(A2)
C      PAUSE'PAUSE AFTER A1,A2,Q(A1,Q(A2): ITERATION FINISHED!!!'
 1000 CONTINUE
      RETURN
      END
```

Appendix C - 4
Convection-Diffusion Model

PROGRAM CONDIF

```
      DIMENSION    C(4,501),Calv(4,501),
     * C0(101),Cso(101),Qsac(101),Q(151),
     * H(101),F(101),THETA1(101),THETA2(101),
     * G(101),ALPHA1(101),ALPHA2(101),BETA1(101),BETA2(101),
     * B1(101),B2(101),B3(101),B4(101)

      COMMON AVTO,A0,DT,DTH,DZ,TIDVOL,VLT,TL,TD,A(101),
     * AV(101),VZ(101),AVZ(201),FA(201),FAZ(201),VOW,CC(151),
     * FLOW(501),VOLIN(501),Z(4,501),Rbeg(4),RV0(4),RV(4),
     * IMAX1,TV,AWL(101),AWD(101),Aso(101),V(151),Falv(151),
     * Vsa(151),SD,Ctrach0

6     FORMAT (10X,8E13.5)

      IMAX=101        !Number of discrete points on the gen coordinate sys
      IMAX1=IMAX-1

      CALL FLOWW(KMAX,KEINS,VSTART)
      CALL GEOM(VSTART,IMAX)

C     INITIALIZE VARIABLES

      DO 20 I=1,IMAX
      C(L,I)=0.78
      C0(0)=0.78
20    Calv(L,I)=0.78
      Ctrach0=0.78

      Cin=0.0                  !Concentration of N2 in the inspired gas
      Dm=0.25                  !Molecular diffusion coefficient for N2
      FLUX=0.0
      CMOUTH=1.0
      DZS=DZ**2
      DZH=DZ/2.
      T=0.0
      U=0
      P=DZS/DT

      DO 100 I=1,IMAX
      H(I)=A(I)/AWL(I)
      F(I)=A(I)*AWL(I)
      THETA1(I)=2*Aso(I)*Dm*((A(I)/2+SD)*Vsa(I))
100   THETA2(I)=DT/(2.0*Vsa(I))


      DO 2000 K=2,KMAX

      T=T+DT
      CMOUTH0=CMOUTH
      Q0=FLOW(K-1)
      DO 150 L=1,4
      C(L,1)=0.0
      IF (VOLIN(K).GT.TV) C(L,1)=Cin

150   CONTINUE
      D1= 0.37*ABS(FLOW(K))*AWD(1)/VZ(1)/DZ
      IF (Q0.GT.0.0.AND.K.GE.KEINS)
     *          CALL ENTRAN(K,KEINS,Q0,CMOUTH,C(L,1),FLUX)

      DO 1000 L=1,4
```

Program CONDIF.FOR

```
C  S IS THE LINEAR SCALING FACTOR BETWEEN THE PREENT VOLUME OF REGION L
C  AND WEIBEL'S ANATOMICAL LUNG MODEL DATA.

      IF(K.GT.2) GOTO 170
      S=(Rbeg(L)/VOW)**(1./3.)
      S2=S**2
      S3=S**3
 170  CONTINUE

      S0=S
      S02=S2
      S03=S3

      Vreg=RVO(L)+.5*(Z(L,K-1)*FLOW(K-1)*Z(L,K)*FLOW(K))
      S=((4.0*Vreg)/VOW)**(1./3.)
      S2=S**2
      S3=S**3

      DO 200 I=1,IMAX1
      CO(I)=C(L,I)
      Cso(I)=(CO(I)+Calv(L,I))/2.0
      Qsac(I)=Z(L,K)*FLOW(K)*(1.-Falv(I))
      Q(I)=Z(L,K)*(1- FA(I))
 200  CONTINUE

       DO 300 I=2,IMAX1
       G(I)= S*(Dm*H(I-1)+Dm*H(I))/2.0
       ALPHA1(I)=S0*Cso(L)/(S3-THETA2(I)*Qsac(I)+S*THETA1(I))
       ALPHA2(I)=(THETA2(I)*Qsac(I)-S*THETA1(I))/
     *     (S3-THETA2(I)*Qsac(I)+S*THETA1(I))
       BETA1(I)=Vsa(I)/DZ*(S3*ALPHA1(I)-S03*Cso(I))
 300   BETA2(I)=Vsa(I)/DZ*(S3*ALPHA2(I))

C   CALCULATE THE COEFFICIENTS TO THE DIFFERENTIAL EQ

      DO 400 I=2,IMAX1
      B1(I)=-(G(I)+G(I-1))/2.0-DZH*Q(I-1)
      B2(I)=P*(S3*F(I)+BETA2(I))+(G(I)+G(I+1))/2+(G(I-1)+G(I))/2.0
      B3(I)=-(G(I)+G(I+1))/2.0+DZH*Q(I+1)
 400  B4(I)=P*(S0*F(I)*CO(I)-BETA1(I))

C  SET THE BOUNDARY CONDITIONS AN I=1 AND AT I=IMAX

      B1(1)=0.0
      B2(1)=1.0
      B4(1)=C(L,1)
      IF (QO.LT.0.) B1(1)=1.0
      IF (QO.LT.0.) B4(1)=FLUX/D1
      B2(IMAX)=1.0
      B3(IMAX)=1.0
      B4(IMAX)=0.0

C  SOLVE THE TRIDIAGONAL MATRIX TO GET THE CONCENTRATIONS ALONG THE
C  THE AIRWAY WITHIN THIS REGION (C(L))

       DO 500 I=1,IMAX
 500   CC(I)=C(L,I)

       CALL SOLVE(B1,B2,B3,B4,IMAX,CC)

      DO 600 I=1,IMAX
 600  C(L,I)=CC(I)

      IF (QO.GT.0.0) CALL ENTRAN(K,KEINS,QO,CMOUTH,C(L,1),FLUX)
 1000  CONTINUE
```

Program CONDIF.FOR

```
 2000  CONTINUE
       STOP
       END
C-----------------------------------------------------------------
      SUBROUTINE GEOM(VSTART,IMAX)

C THIS SUBROUTINE CALCULATES THE ANATOMICAL VARIABLES THAT ARE USED IN
C          THE CONVECTION/DIFFUSION MODEL

       COMMON AVTO,AO,DT,DTH,DZ,TIDVOL,VLT,TL,TD,A(101),
     * AV(101),VZ(101),AVZ(201),FA(201),FAZ(201),VOW,CC(151),
     * FLOW(501),VOLIN(501),Z(4,501),Rbeg(4),RVO(4),RV(4),
     * IMAX1,TV,AWL(101),AWD(101),Aso(101),V(151),Falv(151),
     * Vsa(151),SD,Ctrach0

      DIMENSION Valv(151),Svalv(151),SvalvZ(151),Sawl(151),SawlZ(151)

C   INPUT THE ANATOMICAL DATA

      TDW=1.8
      TLW=22.0

       READ(5,1) (AWL(N),N=1,23)
       READ(5,1) (AWD(N),N=1,23)
  1    FORMAT(13F6.3)
  2    FORMAT(7F11.0)

       DO 10 N=1,16
 10    Valv(N)=0.0

       READ(5,1) (Valv(N),N=17,23)!Alveolar volume per generation

       SD=0.032                !Sac Depth in cm
       Valunit=0.00002671      !Volume of an ALveolar UNIT cm3
       Asop= 0.00052           !Area of a Sac OPening

C  INTERPOLATE THE ANATOMICAL DATA FROM THE 23 GERNERATION GRID
C          TO A FINER GRID OF IMAX POINTS.

      FIMAX1=IMAX1
      DZ=23./FIMAX1

      Aso(1)=0.0
      SValv(1)=0.0
      SVcaw=0.0
      V(1)=0.0
      A(1)=0.0
      Sawl(1)=0.0

      DO 20 N=2,24
      Sawl(N)=Sawl(N-1)+AWL(N)!Summed lungth of conductin airways
      FNUM=(2**N)/4.0         !# of airways/gen in a region
      A(N)=FNUM*3.1416*(AWD(N)**2)/4.!Cross-sectional area of con aw
      SVcaw=SVcaw+A(N)*AWL(N) !Summed vol of con airways
      SValv(N)=SValv(N-1)+Valv(N)!Summed vol of alveoli
 20   V(N)=SValv(N)+SVcaw     !Summed vol of alveoli and airways

      DO 25 N=1,24
      Falv(N)=Valv(N)/V(24)   !Alveolar vol/gen as a fraction of tot vol
 25   FA(N)=V(N)/V(24)        !Fraction of the total volume

      CALL DERIV(FA,24,1.0,FAZ)
      CALL DERIV(SValv,24,1.0,SValvZ)
      CALL DERIV(Sawl,24,1.0,SawlZ)
      CALL DERIV(V,24,1.0,VZ)
```

```
      DO 28 N=1,24
      IF (N.LT.17) SValvZ(N)=0.0
28    IF (FA(N).GT.1.0) FA(N)=1.0

      DO 50 N=2,24
      Nsac=SValvZ(N)/(4.0*Valunit) !# of sacs per generation in a region
      Aso(N)=Asop*Nsac          !Regional Area of sac openings/generation
50    A(N)=VZ(N)/SawlZ(N)

      CALL INTERP(FA,24,1.0,IMAX,DZ)
      CALL INTERP(FAZ,24,1.0,IMAX,DZ)
      CALL INTERP(SValvZ,24,1.0,IMAX,DZ)
      CALL INTERP(SawlZ,24,1.0,IMAX,DZ)
      CALL INTERP(VZ,24,1.0,IMAX,DZ)
      CALL INTERP(A,24,1.0,IMAX,DZ)
      CALL INTERP(V,24,1.0,IMAX,DZ)

C    SCALE WEIBEL'S DATA TO SUBJECT'S LUNG STARTING LUNG VOLUME

      VOW=3545.
      SCALE=(VSTART/VOW)**(1./3.)
C
C    CALCULATE THE NEW TRACHEA GEOMETRIES
C
      TD=TDW*SCALE
      TL=TLW*SCALE
      AO=3.1416*TD**2/4.
      TV=TL*AO

99    FORMAT (11E11.3)
      RETURN
      END
C-----------------------------------------------------------------
      SUBROUTINE FLOWW(KMAX,KEINS,VSTART)

C   THIS SUBROUTINE READS IN THE FLOW AT THE MOUTH, AND THE REGIONAL
C   FLOWS AND VOLUMES THAT WERE CALCULATED IN THE VENTILATION MODEL

        COMMON AVTO,AO,DT,DTH,DZ,TIDVOL,VLT,TL,TD,A(101),
     * AV(101),VZ(101),AVZ(201),FA(201),FAZ(201),VOW,CC(151),
     * FLOW(501),VOLIN(501),Z(4,501),Rbeg(4),RVO(4),RV(4),
     * IMAX1,TV,AWL(101),AWD(101),Aso(101),V(151),Falv(151),
     * Vsa(151),SD,Ctrach0

      INTEGER CRT
      CRT=0
 1    FORMAT(I3,F7.2,F7.1)
 2    FORMAT(9F8.1)
 3    FORMAT(3X,I5)

C   READ THE FLOW DATA FROM THE DATA FILE CALLED FLOW.DAT
C        JMAX = # OF FLOW DATA POINTS FROM EXPERIMENTAL RECORD
C        VSTART = GAS VOLUME OF LUNG AT START OF INHALATION
C        DTC = TIME INCREMENT FOR MEASURED FLOW DATA
C        TIDVOL = MEASURED TIDAL VOLUME
C        RV(L)= REGIONAL VOLUMES FOR REGIONS L=1,4
C        FLOW(J) = MEASURED FLOWS AT THE MOUTH
C        Z(L,J) = FRACTION OF FLOW GOING TO EACH REGION L AT TIME J

C     INPUT FLOW DATA
C
      NI5=5
C   OPEN INPUT FILE 'EXPFLOW.DAT'
      OPEN(NI5,FILE='EXPFLOW.DAT')
      READ(NI5,1) JMAX,DTC,TIDVOL
```

Program CONDIF.FOR

```
      READ(NI5,2) (RV(L),L=1,4)

      VSTART=RV(1)+RV(2)+RV(3)+RV(4)
      DO 40 L=1,4
 40   Rbeg(L)=RV(L)      !Set beginning volumes for each Region L

      READ(NI5,2) (FLOW(J),J=1,JMAX)
      DO 50 L=1,4
 50   READ(NI5,2) (Z(L,J),J=1,JMAX)

C      WRITE(CRT,1) JMAX,VSTART,DTC,TIDVOL
C      WRITE(CRT,2) (FLOW(J),J=1,JMAX)
C      PAUSE'PAUSE TO LOOK AT JMAX,VSTART,DTC,TIDVOL AND FLOWS'
C
C      DETERMINE WHERE THE MEASUED DATA BECOMES NEGATIVE, JEINS

      J=0
 100  J=J+1
      IF(FLOW(J).GE.0.0) GOTO 100
      JEINS=J
C
C   INTERPRET FLOW DATA TO A FINER KMAX GRID.  THE TIME STEP FOR THIS
C   IS BASED ON AN AVERAGE VOLUME INCREMENT OF DELVOL DURING THE INHALATION
C   PART OF THE BREATH.

      JMAX1=JMAX-1
      TMAX=DTC*JMAX1
      DELVOL=20.0  !average volume increment in cc's for time step DT
      TIMEINH = (JEINS-1)*DTC
      DT=TIMEINH*(DELVOL/TIDVOL)
      KMAX1=TMAX/DT
      DT=TMAX/KMAX1
      KMAX=KMAX1+1
      DTH=DT/2.0

      CALL INTERP(FLOW,JMAX,DTC,KMAX,DT)

      WRITE(CRT,2) (FLOW(I),I=1,KMAX)
      WRITE(CRT,2) DTC,DT
      PAUSE'PAUSE TO LOOK AT FLOWS AND DTC AND DT IN SUB FLOWW'
C
C DETERMINE WHEN THE FLOW FIRST BECOMES NEGATIVE, THE END OF INHALATION.
C         LOOK FOR KCHECK FLOWS LESS THAN ZERO.
C
      KCHECK=3
      KC=0
      K=0
 150  K=K+1
C      WRITE(CRT,3)K
      IF(FLOW(K).GT.0) GOTO 200
      KC=KC+1
      IF(KC.EQ.KCHECK) GOTO 250
      GOTO 150
 200  KC=0
      GOTO 150
 250  KEINS=K-(KCHECK-1)
C      PAUSE'PAUSE TO LOOK AT ALL THE Xs'

C  INTEGRATE FLOWS TO GET TIDAL VOLUME BASED ON FLOW DATA (FTIDVOL)

      KEINS1=KEINS-1
      FTIDVOL=0.0
      DO 300 K=1,KEINS1
 300  FTIDVOL=FTIDVOL+DTH*(FLOW(K)+FLOW(K+1))
```

Program CONDIF.FOR

```
C     NORMALIZE THE FLOWS WITH RESPECT TO THE MEASURED TIDAL VOLUME

      SCALE=TIDVOL/FTIDVOL
      DO 400 K=1,KMAX
 400  FLOW(K)=SCALE*FLOW(K)

C    CALCULATE THE INHALD VLUME (VOLIN) BASED ON THE SCALED FLOWS

      VOLIN(1)=0.0
      DO 500 K=2,KMAX
 500  VOLIN(K)=VOLIN(K-1)+DTH*(FLOW(K-1)+FLOW(K))

C      WRITE(CRT,2) (FLOW(K),K=1,KMAX)
C      WRITE(CRT,2) (VOLIN(K),K=1,KMAX)
C      PAUSE'PAUSE TO LOOK AT FLOW(K) AND VOLIN(K) IN SUB FLOWW'

      RETURN
      END
C-----------------------------------------------------------------------
      SUBROUTINE ENTRAN(K,KEINS,Q0,CMOUTH,CTRACH,FLUX)
         COMMON AVTO,A0,DT,DTH,DZ,TIDVOL,VLT,TL,TD,A(101),
     * AV(101),VZ(101),AVZ(201),FA(201),FAZ(201),VOW,CC(151),
     * FLOW(501),VOLIN(501),Z(4,501),Rbeg(4),RVO(4),VR(4),
     * IMAX1,TV,AWL(101),AWD(101),Aso(101),V(151),Falv(151),
     * Vsa(151),SD,Ctrach0

      DIMENSION C(201),CO(201),SDC(201)

      AQ0=ABS(FLOW(K))
      RET=AQ0*TD/A0/0.15
      FK=0.05/(2.14*ALOG(RET)-3.6)
      D=FK*TD*AQ0/A0
      IMAX=21
      IMAX1=IMAX-1
      FIMAX1=IMAX1
      DX=TL/FIMAX1
      NT=1.+2.*D*DT/DX**2
      FNT=NT
      L=0.5+(AQ0/A0)*DT/FNT/DX
      FL=L
      DX=(AQ0/A0)*(DT/FNT)/FL
      DX2=DX**2
      IMAX=TL/DX+1.5
      IMAX1=IMAX-1
      P=D*(DT/FNT)/DX**2
      WRITE (6,18) NT,IMAX,L,DX,D,P,CMOUTH,CTRACH
18    FORMAT (5X,3I5,8E14.3)
      L1=L+1
      IML=IMAX-L
      IML1=IML+1
      NUMJ=KEINS
      DO 15 I=1,IMAX
15    C(I)=0.0
20    CONTINUE
      IQ0=Q0/AQ0
      NCYC=1+(K-1)/NUMJ
      JCYC1=1+(NCYC-1)+NUMJ
      IF (K.EQ.JCYC1) C(1)=(1.+C(1))/2.
      CTRCHO=C(IMAX)
      DO 80 N=1,NT
      DO 25 I=1,IMAX
25    CO(I)=C(I)
      DO 30 I=2,IMAX1
30    SDC(I)=CO(I-1)-2.*CO(I)+CO(I+1)
      SDC(1)=1.-2.*CO(1)+CO(2)
```

Program CONDIF.FOR

```
      SDC(IMAX)=SDC(IMAX1)
      IF (IQ0.LT.0) GO TO 40
      DO 32 I=1,L
32    C(I)=1.0
      DO 35 I=L1,IMAX
      K=I-L
35    C(I)=C0(K)+P*SDC(K)
      CTRACH=C(IMAX)
      GO TO 60
40    CONTINUE
      C(IMAX)=Ctrach0 +N*(CTRACH-Ctrach0)/FNT
      DO 50 I=1,INL
      K=I+L
50    C(I)=C0(K)+P*SDC(K)
      DO 55 I=INL1,IMAX
      F=(I-INL)/FL
55    C(I)=C0(IMAX)+F*(C(IMAX)-C0(IMAX))
60    CONTINUE
80    CONTINUE
      FLUX=D*(C(IMAX)-C(IMAX-1))/DX
      CMOUTH=C(1)
90    FORMAT (10E11.3)
100   CONTINUE
      RETURN
      END
C----------------------------------------------------------------
      SUBROUTINE INTERP(Y,IMAX1,DX1,IMAX2,DX2)

      DIMENSION Y(201),X1(201),X2(201),Y2(201)

      DO 10 I1=1,IMAX1
10    X1(I1)=(I1-1)*DX1
      DO 20 I2=1,IMAX2
20    X2(I2)=(I2-1)*DX2
      DO 30 I2=1,IMAX2
      I1=X2(I2)/DX1+DX1/2.
      I1=I1+1
      IF (I1.EQ.1) I1=2
      IF (I1.EQ.IMAX1) I1=IMAX1-1
      U=(X2(I2)-X1(I1))/DX1
      D1=Y(I1+1)-Y(I1-1)
      D2=Y(I1+1)-2.*Y(I1)+Y(I1-1)
30    Y2(I2)=Y(I1)+.5*D1*U+.5*D2*U**2
      DO 40 I2=1,IMAX2
40    Y(I2)=Y2(I2)
      RETURN
      END
C----------------------------------------------------------------
      SUBROUTINE SOLVE(A,B,C,D,NUM,U)
      DIMENSION A(201),B(201),C(201),D(201),U(201)
      A(1)=A(1)/B(1)
      D(1)=D(1)/B(1)
      A(NUM)=0.0
      U(NUM+1)=0.
      DO 10 N=2,NUM
      A(N)=A(N)/(B(N)-C(N)*A(N-1))
10    D(N)=(D(N)+C(N)*D(N-1))/(B(N)-C(N)*A(N-1))
      DO 20 N=1,NUM
      K=NUM+1-N
20.   U(K)=A(K)*U(K+1)+D(K)
      RETURN
      END
C----------------------------------------------------------------
       SUBROUTINE DERIV(FUN,NUM,H,DFUN)
       DIMENSION FUN(201),DFUN(201)
```

Program CONDIF.FOR

```
      N=NUM-2
      DFUN(1)=(-25.0*FUN(1)/12.0+4.0*FUN(2)-3.0*FUN(3)+4.0*FUN(4)/3.0
     *          -FUN(5)/4.0)/H
      DFUN(2)=(-FUN(1)/4.0-5.0*FUN(2)/6.0+3.0*FUN(3)/2.0-FUN(4)/2.0
     *          +FUN(5)/12.0)/H
      DO 10 N=3,N
10    DFUN(N+2)=(FUN(N-2)/12.0-2.0*FUN(N-1)/3.0+2.0*FUN(N+1)/3.0
     *               -FUN(N+2)/12.0)/H
      DFUN(NUM-1)=(FUN(NUM)/4.0+5.0*FUN(NUM-1)/6.0-3.0*FUN(NUM-2)/2.0
     *                   +FUN(NUM-3)/2.0-FUN(NUM-4)/12.0)/H

      DFUN(NUM)=(25.0*FUN(NUM)/12.0-4.0*FUN(NUM-1)+3.0*FUN(NUM-2)
     *                 -4.0*FUN(NUM-3)/3.0+FUN(NUM-4)/4.0)/H

      RETURN
      END
```

APPENDIX D

Cardiovascular Computer Programs

Appendix D
Cardiovascular Computer Programs

PROGRAM CVMODEL

```
C...
C...  PROGRAM CVMODEL CALLS: (1) SUBROUTINE INITAL TO DEFINE THE ODE
C...  INITAL CONDITIONS, (2) SUBROUTINE RKF45 TO INTEGRATE THE ODES,
C...  AND (3) SUBROUTINE PRINT TO PRINT THE SOLUTION.
C...
C...  THE FOLLOWING CODING IS FOR 500 ODES.  IF MORE ODES ARE TO BE INTE-
C...  GRATED, ALL OF THE 500'S SHOULD BE CHANGED TO THE REQUIRED NUMBER
      IMPLICIT DOUBLE PRECISION (A-H), DOUBLE PRECISION (O-Z)
      INTEGER NI, NO, NEQN, NSTOP, NORUN
      COMMON/T/          T,      NSTOP,      NORUN,    PP,   TIM
     1      /Y/     Y(500)
     2      /F/     F(500)
C...
C...  THE NUMBER OF DIFFERENTIAL EQUATIONS IS IN COMMON/N/ FOR USE IN
C...  SUBROUTINE FCN
      COMMON/N/      NEQN  ! TWO EQUATIONS PER VASCULAR SEGMENT
C...
C...  COMMON AREA TO PROVIDE THE INPUT/OUTPUT UNIT NUMBERS TO OTHER
C...  SUBROUTINES
      COMMON/IO/       NI,        NO
C...
C...  ABSOLUTE DIMENSIONING OF THE ARRAYS REQUIRED BY RKF45
C...
C...    THE USER MUST PROVIDE STORAGE IN HIS CALLING PROGRAM FOR THE ARRAYS
C...    IN THE CALL LIST  -      Y(NEQN) , WORK(3+6*NEQN) , IWORK(5)  ,
C...    DECLARE F IN AN EXTERNAL STATEMENT, SUPPLY SUBROUTINE F(T,Y,YP) AND

      DOUBLE PRECISION YV(500), WORK(3500)
      INTEGER IWORK(5)
C...
C...  EXTERNAL THE DERIVATIVE ROUTINE CALLED BY RKF45
C...
      EXTERNAL FCN
C...
C...  ARRAY FOR THE TITLE (FIRST LINE OF DATA), CHARACTERS  END OF RUNS
      CHARACTER TITLE(20)*4, ENDRUN(3)*4
C...
C...  DEFINE THE CHARACTERS  END OF RUNS
      DATA ENDRUN/'END ','OF R','UNS '/
C...
C...  DEFINE THE INPUT/OUTPUT UNIT NUMBERS
      NI=5
      NO=6
C...
C...  OPEN INPUT AND OUTPUT FILES
      OPEN(NI,FILE='CVDATA.DAT')
      OPEN(NO,FILE='CVOPUT.TXT',BLOCKSIZE=2048)
C...
C...  INITIALIZE THE RUN COUNTER
      NORUN=0
C...
C...  BEGIN A RUN
1     NORUN=NORUN+1
C...
C...  INITIALIZE THE RUN TERMINATION VARIABLE
      NSTOP=0
C...
C...  READ THE FIRST LINE OF DATA
C...
      READ(NI,1000,END=999) (TITLE(I), I = 1, 20)
```

Program CVMODEL.FOR

```
C...
C...  TEST FOR  END OF RUNS  IN THE DATA
C...
      DO 2 I = 1, 3
      IF(TITLE(I) .NE. ENDRUN(I)) GO TO 3
2     CONTINUE
C...
C...  AN END OF RUNS HAS BEEN READ, SO TERMINATE EXECUTION
999   STOP
C...
C...  READ THE SECOND LINE OF DATA
C...
3     READ(NI,*,END=999) TO, TF, TP
C...
C...  READ THE THIRD LINE OF DATA
C...
      READ(NI,*,END=999) NEQN, ERROR
C...
C...  PRINT A DATA SUMMARY
      WRITE(NO,1003)NORUN,(TITLE(I), I = 1, 20),
     1                     TO, TF, TP,
     2                     NEQN, ERROR
      WRITE(*,1003) NORUN, (TITLE(I), I = 1, 20),
     1                     TO, TF, TP,
     2                     NEQN, ERROR
C...
C...  INITIALIZE TIME
      T = TO
C...
C...  SET THE INITIAL CONDITIONS
      CALL INITAL
C...
C...  SET THE INITIAL DERIVATIVES (FOR POSSIBLE PRINTING)
      CALL DERV
C...
C...  PRINT THE INITIAL CONDITIONS
      CALL PRINT(NI, NO)
C...
C...  SET THE INITIAL CONDITIONS FOR SUBROUTINE RKF45
      TV = TO
      DO 5 I = 1, NEQN
      YV(I) = Y(I)
5     CONTINUE
C...
C...  SET THE PARAMETERS FOR SUBROUTINE RKF45
C...
C...   FIRST CALL TO RKF45
C...
      RELERR = ERROR
      ABSERR = ERROR
      IFLAG = 1
      TOUT = TO + TP
C...
C...  CALL SUBROUTINE RKF45 TO START THE SOLUTION FROM THE INITIAL
C...  CONDITION (IFLAG = 1) OR COMPUTE THE SOLUTION TO THE NEXT PRINT
C...  POINT (IFLAG = 2)
C...
4     CALL RKF45(FCN,NEQN,YV,TV,TOUT,RELERR,ABSERR,IFLAG,WORK,IWORK)
C...
C...  PRINT THE SOLUTION AT THE NEXT PRINT POINT
C...
      T=TV
      TOUT = TV + TP
      PRINT *,"Time = ", T
      DO 6 I = 1, NEQN
```

```
          Y(I) = YV(I)
6      CONTINUE
       CALL DERV
       CALL PRINT(NI,NO)
C...
C...   TEST FOR AN ERROR CONDITION
       IF(IFLAG .NE. 2) THEN
C...
C...      PRINT A MESSAGE INDICATING AN ERROR CONDITION
          WRITE(NO,1004) IFLAG
C...
C...      GO ON TO THE NEXT RUN
          GO TO 1
       END IF
C...
C...   CHECK FOR A RUN TERMINATION
       IF(NSTOP .NE. 0) GO TO 1
C...
C...   CHECK FOR THE END OF THE RUN
C...
       IF(TV .LT. (TF - 0.5D0*TP)) GO TO 4
C...
C...   THE CURRENT RUN IS COMPLETE, SO GO ON TO THE NEXT RUN
       GO TO 1
C...
C...   ******************************************************************
C...
C...   FORMATS
C...
1000   FORMAT(20A4)
1001   FORMAT(3E10.0)
1002   FORMAT(I5,20X,E10.0)
1003   FORMAT(1H1,
     1 ' RUN NO. - ',I3,2X,20A4,//,
     2 ' INITIAL T - ',E10.3,//,
     3 '   FINAL T - ',E10.3,//,
     4 '   PRINT T - ',E10.3,//,
     5 ' NUMBER OF DIFFERENTIAL EQUATIONS - ',I5,//,
     6 ' MAXIMUM INTEGRATION ERROR - ',E10.3,//,
     7 1H1)
1004   FORMAT(1H ,//,' IFLAG = ',I3,//,
     1 ' INDICATING AN INTEGRATION ERROR, SO THE CURRENT RUN'      ,/,
     2 ' IS TERMINATED.  PLEASE REFER TO THE DOCUMENTATION FOR'    ,/,
     3 ' SUBROUTINE',//,25X,'RKF45',//,
     4 ' FOR AN EXPLANATION OF THESE ERROR INDICATORS'              )
       END
       SUBROUTINE FCN(TV,YV,YDOT)
C...
C...   SUBROUTINE FCN IS AN INTERFACE ROUTINE BETWEEN SUBROUTINES RKF45
C...   AND DERV
C...
C...   NOTE THAT THE SIZE OF ARRAYS Y AND F IN THE FOLLOWING COMMON AREA
C...   IS ACTUALLY SET BY THE CORRESPONDING COMMON STATEMENT IN MAIN
C...   PROGRAM HEADHIT
       IMPLICIT DOUBLE PRECISION (A-H), DOUBLE PRECISION (O-Z)
       INTEGER NEQN, NSTOP, NORUN
       COMMON/T/          T,      NSTOP,     NORUN
     1       /Y/       Y(500)
     2       /F/       F(500)
C...
C...   THE NUMBER OF DIFFERENTIAL EQUATIONS IS AVAILABLE THROUGH COMMON
C...   /N/
C...
       COMMON/N/       NEQN
C...
```

Program CVMODEL.FOR

```
C...  ABSOLUTE DIMENSION THE DEPENDENT VARIABLE, DERIVATIVE VECTORS
      DOUBLE PRECISION YV(500), YDOT(500)
C...
C...  TRANSFER THE INDEPENDENT VARIABLE, DEPENDENT VARIABLE VECTOR
C...  FOR USE IN SUBROUTINE DERV
C...
      T = TV
      DO 1 I = 1, NEQN
         Y(I) = YV(I)
1     CONTINUE
C...
C...  EVALUATE THE DERIVATIVE VECTOR
C...
      CALL DERV
C...
C...  TRANSFER THE DERIVATIVE VECTOR FOR USE BY SUBROUTINE RKF45
C...
      DO 2 I = 1, NEQN
         YDOT(I) = F(I)
2     CONTINUE
      RETURN
      END
```

Program CVSUBS.FOR

DECK CVSUBS.FOR - SUBROUTINES REQUIRED TO IMPLEMENT A DYNAMIC MODEL OF THE HUMAN

```
C...  CARDIOVASCULAR SYSTEM
C...
C...  LAST REVISION:  1/22/94
C...
      SUBROUTINE INITIAL
C...
C...  THE model described herein parallels the development presented in a paper by
C...  White, RJ, Croston, RC and Fitzjerrell, DG.  Cardiovascular Modelling:  Simulating
C...  the Human Cardiovascular Response to Exercise, Lower Body Negative Pressure, Zero
C...  Gravity and Clinical Conditions.  Adv. Cardiovasc. Phys. (Part I), pp. 195-229 (Karger,
C...  Basel 1983).  It also draws from papers by Jaron, et al who took a similar approach
C..   In particular many of the parameter values for the physical properties of the segments
C...  were taken from:
C...
C...       Jaron, D, Moore, TW, and Bai, J.  Cardiovascular Response to Acceleration Stress:
C...       A Computer Simulation.  Proceedings of the IEEE, Vol 76, No 6, pp. 700-707 (1988).
C...
C...  However, some of the parameters listed in Jaron were clearly in error.  Where new
C...  parameters were required they were derived to yield generally acceptable
C...  flow/pressure/volume and compliance characteristics of the various
C...  cardiovascular subdivisions.  In particular, data from
C...
C...       Burton, Alan, C.  Physiology and Biophysics of the Circulation.  Year
C...       Book Medical Publishers, Inc.  Chicago, IL (1965)
C...
C...     and,
C...
C...       Guyton, A.C.  Textbook of Medical Physiology.  7th Ed.,
C...       W.B. Saunders, Philadelphia, PA. (1985).
C...
C...  The model describes the spatial and temporal variation in the mean
C...  blood pressure along the z-axis of the body.  The model neglects the
C...  non-linear and convective terms in the Navier-Stokes Equation.  The
C...  model also assumes negligible radial flow.  The flow is assumed laminar
C...  except in the ascending an descending aorta where fluid flow resis-
C...  tance multiplied by 33 to account for turbulent pressure losses.
C...
C...  NOTE:  THE SUBSCRIPT NOTATION INDICATES PARTIAL DERIVATIVE WRT THE
C...         SUBSCRIPT  E.G.  Xt ==> THE FIRST PARTIAL OF X WRT TIME
C...         IN THIS MODEL:
C...
C...                   t = Time                          [sec]
C...                   r = Radius of vascular segment    [m]
C...                   l = Length of vascular segment    [m]
C...                rho0 = density of blood              [kg/m^3]
C...                 mu0 = viscosity of blood            [N-sec/m^2]
C...
C... MODEL FOR ARTERIAL SEGMENTS
C...
C...  The following set of simultaneous equations are solved f    ach
C...  arterial vascular segment.
C...
C...    Pt(t) = 1/C*(Qin(t) - Qout(t)) + R2*(Qtin -Qtcut)
C...    Qt(t) = 1/L*(Pin(t) - Pout(t) + PGz - R*Q(t))
C...    rt(t) = 1/(2*Pi*r*l)*(Qin(t) - Qout(t))
C...
C...    Where,
C...
C...     P = The pressure in the segment               [Pa]
C...     Q = The segmental volume flow                 [m^3/sec]
C...     C = The capacitance of the segment            [m^3/Pa]
```

Program CVSUBS.FOR

```
C...     L = The inertance of the segment               [kg/m^4] or [Pa-sec^2/m^3]
C...     Ra = The viscous flow resistance in the segment  [m].
C...     PGz = The hydrostatic pressure difference
C...            across the segment because of gravity    [Pa]
C...
C...
C...    And, the following approximations for Ra, La, and Ca are taken from
C...    a paper by
C...
C...       Rideout, et al.  Difference-Differential Equations for Fluid
C...       Flow in Distensible Tubes.  IEEE Transactions on Bio-Medical
C...       Engineering.  Vol BME-14, NO. 3, pp 171-177. Jul 1967.
C...
C...       Ra = 81*mu0*l/(8*pi*r^4)
C...       La =  9*rho0*l^2/(4*VOL)
C...       Ca =  3*r*VOL*l/(2*E*h)
C...
C...    Where,
C...
C...       E = Young's modulus for vessel wall            [Pa]
C...       h = Vessel Wall thickness                      [m].
C...
C...    Finally,
C...
C...       PGz = rho0*Gz*g0*l*cos(theta).
C...
C...    Where,
C...
C...       Gz    = The z-axis "G-level" in units of earth's gravity   [unitless]
C...       g0    = The earth's gravitational acceleration             [m/sec^2]
C...       theta = The angle between the segment and the z-axis       [radians].
C...
C...
C... MODEL FOR VENOUS SEGMENTS
C...
C...    The model for venous segments was adapted from
C...
C...       Snyder, et al.  Computer Simulation Studies of
C...       Venous Circulation.  IEEE Trans Bio-Med Engr Vol BME-16,
C...       NO. 4 pp 325-334. Oct 1969.
C...
C...    The unstressed internal volume of a vascular segment in assumed to be
C...
C...        V = Pi*r^2*l                         [m^3].
C...
C...    When the contained volume, v, is greater than V, the transmural
C...    pressure is assumed to be related to the contained volume by,
C...
C...        dPwall = 1/C*v.
C...
C...    Where C is the vascular compliance as defined above.  For v < V,
C...
C...        dPwall = 1/(20*C)*v.
C...
C...    In a collapsed or partially collapsed vein the flow-pressure
C...    relationship based on an (assumed) elliptical cross-section
C...    and is given by
C...
C...        Qt = 1/Lv*(Pin(t) - Pout(t) + PGz - Rv*Q(t))
C...
C...    Where,
C...
C...        Lv =  9*rho0*l^2/(4*v), and
C...
C...             | 81*mu0*pi^2*l^4*r^2/(8*v^3) for v < V
```

```
C...       Rv = | or
C...            | 81*mu0*l/(8*pi*r^4)    for v > or = V
C...            | or 81*mu0*l^3/(8*v^2)      ".
C...
C...
C...
C...
C...   PERIPHERAL CAPILLARY BEDS.
C...
C...     Peripheral capillary resistance and capacitance
C...     are modelled as lumped parameter models ("T" circuits)
C...     following the method of Jaron, et al (cited above).
C...
C...
C...
C...                      Ra         Rv
C...             Pa  o--/\/\/\--|--/\/\/\---o Pv
C...                            |
C...                           --- Cperipheral
C...                           -|-
C...                            |
C...                          Pexternal (ie P G-suit)
C...                            |
C...                          Pref (Atmospheric)
C...
C...
C...
C...    PULMONARY CIRCULATION
C...
C...        The pulmonary circulation is modelled as a lumped
C...        parameter model again following Jaron's method.  The
C...        model is a "PI" circuit as shown below.
C...
C...
C...         Prtvent     Rpul
C...      Qvr>---->|---/\/\/\-----| Pleft atrium
C...  venous       |              |
C...  return      --- Cpul        |
C...              -|-             |
C...               |              |
C...          Intra-Thoracic Pressure
C...
C...
C... CARDIAC CIRCULATION/OUTPUT
C...
C...   The chambers of the heart are modelled as variable capacitances
C...   separated by one-way valves.  The pulmonic and aortic valves are
C...   also modelled as one-way valves.  The general method of modelling
C...   The heart and its circulation follows the method of Snyder et al.
C...   Output flow from the left atrium and left ventricle are modelled
C...   as simple half-wave rectified sinusoids whose volume flows are
C...   estimated from pulmonary venous flow.
C...
C...
C... PRESSURE REFERENCE
C...
C...   The pressure reference for the model is located at the tri-cuspid
C...   valve which presumably tracks intrathoracic pressure.
C...
C...
C... MODEL MECHANICS
C...,
C...    For each vessel segment, three coupled non-linear differential
C...    equations must be solved simultaneously.  There are 20 vascular
C...    segments (the pulmonary circuit is segment 1).  There are 10
```

```
C...    capillary bed segments modelled as simple resistance and compliance
C...    circuits which are affected directly by extra-vascular pressure.
C...    The pressures and flows in the various segments are coupled by
C...    their spatial connection.  The following table gives the
C...    approximate anatomical location and the corresponding z-axis
C...    coordinate (measured from the tricuspid valve) for each segment.
C...    The z-axis coordinates were based on a 177 cm tall standing man.
C...
C...                                              Arterial
C...                                              Origin
C... Segment             Anatomical              Z-axis          Peripheral
C... Number              Location                coordinate (cm)   Bed
C...=====================================================================
C...    1             Mid-Pulmonary                 0             X
C...    2             Ascending Aorta               0
C...    3             Descending Aorta              5
C...    4             Thoracic Aorta/Vena Cava     -8             X
C...    5             Diaphragm/Lower Lung        -15
C...    6             Renal/Hepatic               -22             X
C...    7             Splanchnic                  -32             X
C...    8             Buttocks                    -42             X
C...    9             Femoralis                   -50
C...   10             Mid Thigh                   -65             X
C...   11             Knee/Poplateal              - 80
C...   12             Calf                        -100            X
C...   13             Ankle                       -125
C...   14             Foot                        -132            X
C...   15             Aortic Arch                   6
C...   16             Lower Neck                   15
C...   17             Carotid Sinus                25
C...   18             Ophthalmic                   34             X
C...   19             Mid Brain                    37
C...   20             Cerebral                     42             X
C...
C...
C...
C...   Initial Conditions (t = 0)
C...
C...       The initial conditions for pressure, flow, and volume are
C...       set based on a steady-state solution for the model at 1 Gz
C...       for a supine posture.  For other postures, the initial theta's
C...       for the segments must be changed.
C...
C...     Postural and/or Gz changes during a simulation.
C...
C...       This can be most easily accomplished by adding time
C...       varying profiles for Gz and the theta's in SUBROUTINE DERV
C...       which forms the derivatives for CVMODEL.
C...
C...  THE NUMERICAL METHOD OF LINES (W.E. SCHIESSER) IS EMPLOYED TO
C...  INTEGRATE THE COUPLED DIFFERENTIAL EQUATIONS (DES).
C...
C...  ODE COMMON
C...
C...   /Y/ time variables
C...   /F/ time derivatives of variables
C...   /R/ & /I/ real and integer parameters required to define constants and
C...           define the spatial integration grid.
C...
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      PARAMETER (NEQ = 20, NPSEG = 10)
      .INTEGER NSTOP, NORUN, IP
      INTEGER*2 ALIN, ALOUT, VLIN, VLOUT, PVS, PIN, POUT
      DOUBLE PRECISION MM2PA, mu0
      COMMON/T/        T,     NSTOP,     NORUN    ! Run Parameters
```

Program CVSUBS.FOR

```
C...
C...  Arrays for segmental variables Pressure (P), Flow(Q), and radii (r)
C...
     1  /Y/    HP(4),      HQ(4),                   ! Heart's Chambers RA=1
     *         AP(NEQ),   AQ(NEQ),   Ar(NEQ),      ! Arterial P, Q, r
     *         VP(NEQ),   VQ(NEQ),   Vr(NEQ),      ! Venous P, Q, r
     *         VOP(3),     VOQ(3),                  ! Venous flows into heart
     *         PP(NEQ),   PQ(NEQ),                  ! Peripheral P, Qin, Qout
C...
C...  Time derivatives of the segmental variables: Pt, Qt, rt
C...
     2  /F/   HPt(4),     HQt(4),                      ! Heart's Chambers RA=1
     *        APt(NEQ),  AQt(NEQ),    Art(NEQ),       ! Arterial Pt, Qt, rt
     *        VPt(NEQ),  VQt(NEQ),    Vrt(NEQ),       ! Venous Pt, Qt, rt
     *        VOPt(3),    VOQt(3),                     ! Venous flows into heart
     *        PPt(NEQ),  PQt(NEQ),                     ! Peripheral Pt, QINt, QOUTt
C...
C... Parameters necessary to form the differential equations
C...
C...    Pi  -  3.14159...
C...    g0  -  9.80665 [m/sec^2]  earth's acceleration of gravity
C...  rho0  -  1050.   [kg/m^3]   density of whole blood (45% Hct)  | Assumed
C...   mu0  -  2.7     [cp]       viscosity of whole blood (45% Hct)| Constant
C...   D2R  -  pi/180  [radians/degree] scale factor
C...
     3  /R/   Pi, g0, rho0, mu0,  D2R, MM2PA, R2,       ! Constants
     *         ZAO(NEQ), ZAT(NEQ), ZVO(NEQ), ZVT(NEQ),   ! Arterial & Venous
     *         THETA(NEQ), HCAP(4), HVOL(4), HVR(4),    ! Orientation angle
     *         ALO(NEQ),  ArU(NEQ),  AE(NEQ), Ah(NEQ),   ! Arterial l, r, E, h
     *         ACAP(NEQ), ARES(NEQ), AINERT(NEQ),        ! Arterial Capacitance, resistance
     *         VLO(NEQ),  VrO(NEQ),  VrU(NEQ),           ! Venous l, r, rUNSTRESSED
     *        VCAPO(NEQ), VCAP(NEQ), VRES(NEQ), VINERT(NEQ), ! Venous Capacitance, resistance, inertance
     *         PRA(NEQ), PRV(NEQ), PCAP(NEQ),           ! Peripheral Ra, Rv, C
     *         PINERT(NEQ), PVOL(NEQ),                  ! Peripheral I, V
     *         PAO(NEQ),  PVO(NEQ),                     ! Initial P conditions
     *         QAO(NEQ),  QVO(NEQ),                     ! Initial Q conditions
     *         AVOL(NEQ),  VVOL(NEQ),                   ! A & V Volumes
     *         PEXT(NEQ),                               ! Externally applied Pressure
     *         TO, GSTART, GMAX, TBRK1, TBRK2, TMAX, GFIN, ! G-Profile parameters
     *         Gz, ADPG(NEQ), VDPG(NEQ)                 ! Gz & Delta P from G
     4  /I/   IP, NDXPER(NPSEG),                        ! Peripheral Bed Indexes
     *         ALIN(NEQ), ALOUT(NEQ),                   ! Linkage Data arterial
     *         VLIN(NEQ), VLOUT(NEQ),                   ! Venous
     *         PIN(NEQ), POUT(NEQ),                     ! Peripheral
     *         PVS(NEQ),                                ! Number parallel venous segments
     *         IFOOTSEG, IHEADSEG, IHEARTSEG            ! Foot, Head, & Heart seg nums
C...
C...  Note that artery output flow feeds artery input or peripheral bed
C...  input, artery input comes only from arteries or the heart, venous
C...  output flows to veins or the heart, venous input comes from veins
C...  and/or peripheral beds, peripheral beds are fed only by arteries, and
C...  feed only veins.  The index (i) for an segment refers to its input
C...  flow and pressure.  The output pressure for a segment is
C...  stored in P(i+1) and its output flow is stored at Q(i+1)
C...
C...
C...  Define Some Constants and Parameters
C...
C...  Heart valve resistances
C...
      DATA HVR/ 1.48D+6, 1.48D+6, 2.96D+6, 2.96D+6/
C...
C...  Heart chamber capacitances
C...
      DATA HCAP/ 2.25D-7, 6.55D-7, 1.12D-7, 3.28D-7/
```

Program CVSUBS.FOR

```
C...
C...  Arterial Segment length [meters]
C...
      DATA ALO/ 3.D-2,  3.D-2,  5.D-2,  7.D-2,  5.D-2,
     *         12.D-2, 10.D-2,  8.D-2, 15.D-2, 20.D-2,
     *         15.D-2, 25.D-2,  5.D-2,  2.D-2,  9.D-2,
     *         10.D-2,  9.D-2,  3.D-2,  4.D-2,  1.D-2/
C...
C...  Arterial Segment radii at t = 0  [meters]
C...  These radii are based on flow resistance
C...
      DATA ArU /6.000D-3, 1.500D-2, 1.000D-2, 1.200D-2, 1.000D-2,
     *          6.000D-3, 5.500D-3, 4.000D-3, 3.500D-3, 4.000D-3,
     *          2.800D-3, 4.500D-3, 2.750D-3, 1.000D-3, 6.000D-3,
     *          3.000D-3, 3.000D-3, 3.000D-3, 3.000D-3, 1.500D-3/
C...
C...  These radii are based on capacitance and resistance
C...
C...  Young's modulus for arterial segment walls [Pa]
C...
      DATA AE /  2.5D+5,  5.0D+5,  5.0D+5,  7.0D+5,  7.0D+5,
     *           8.0D+5,  8.0D+5,  8.0D+5,  8.0D+5,  1.0D+6,
     *           1.0D+6,  1.0D+6,  1.0D+6,  1.0D+6,  8.0D+5,
     *           8.0D+5,  8.0D+5,  8.0D+5,  8.0D+5,  8.0D+5/
C...
C...  Wall thickness for arterial segments [meters]
C...
      DATA Ah /  2.D-4,  16.D-4, 16.D-4,  14.D-4, 12.D-4,
     *          12.D-4,  10.D-4, 10.D-4,   8.D-4,  8.D-4,
     *           8.D-4,   6.D-4,  6.D-4,   5.D-4,  5.D-4,
     *           6.D-4,   6.D-4,  6.D-4,   5.D-4,  5.D-4/
C...
C...  Number of parallel venous paths in each segment
C...
      DATA PVS/  4,  1,  1,  2,  4,  4,  4,  4,  2,  2,
     *           2,  2,  4,  4,  2,  2,  4,  4,  4,  4/
C...
C...  Venous segment length [meters]
C...
      DATA VLO/ 10.D-2,  2.D-2,  2.D-2, 10.D-2, 20.D-2,
     *          50.D-2, 40.D-2, 32.D-2, 30.D-2, 40.D-2,
     *          30.D-2, 40.D-2, 20.D-2,  8.D-2, 27.D-2,
     *          20.D-2, 36.D-2, 12.D-2, 16.D-2,  4.D-2/
C...
C...  Initial radii for Venous segment [meters]
C...
      DATA VrU/ 5.0D-3,  5.0D-3,  5.0D-3,  3.0D-3,  3.5D-3,
     *          2.5D-3,  4.0D-3,  4.0D-3,  3.0D-3,  3.0D-3,
     *          3.5D-3,  3.3D-3,  2.5D-3,  1.8D-3,  4.5D-3,
     *          4.0D-3,  4.0D-3,  3.3D-3,  3.0D-3,  2.0D-3/
C...
C...   Venous capacitance [m^3/Pa]
C...
      DATA VCAPO/ 5.0D-8,  5.000D-8,  1.000D-8,  5.000D-8,  5.0D-8,
     *            5.0D-8,  5.000D-8,  4.000D-8,  3.000D-8,  2.5D-8,
     *            8.0D-8,  5.000D-8,  5.000D-8,  5.000D-9,  2.0D-8,
     *            5.0D-8,  8.000D-8,  1.000D-8,  1.000D-8,  3.0D-8/
C...
C...  Peripheral Segment Indices (There are 10 peripheral segments).
C...
      DATA NDXPER / 1, 3, 6, 7, 8, 10, 12, 14, 18, 20/
C...
C...  Peripheral Vascular Capacitance [m^3/Pa]
C...
      DATA PCAP/ 1.13D-7,  0.0D000,  3.75D-8,  0.0D00,  0.0D000,
```

```
     *           7.50D-8,  1.13D-7,  7.50D-8,  0.0000,  3.75D-8,
     *           0.00D00,  3.75D-8,  0.00D00,  3.75D-8, 0.00D00,
     *           0.00D00,  0.00D00,  3.75D-8,  0.00D00, 3.75D-8/
C...
C...  Peripheral Vascular Resistance Arterial side [Pa-sec/m^3]
C...  Assign 9.9D99 to segments with no peripheral bed.
C...
      DATA PRA/  1.20D+7,  9.99D99,  1.41D+9,  9.99D99,  9.99D99,
     *           1.39D+9,  3.44D+8,  1.37D+9,  9.99D99,  1.34D+9,
     *           9.99D99,  2.18D+9,  9.99D99,  6.34D+9,  9.99D99,
     *           9.99D99,  9.99D99,  1.32D+10, 9.99D99,  6.67D+8/
C...
C...  Peripheral Vascular Resistance Venous side [Pa-sec/m^3]
C...  Assign 9.9D99 to segments with no peripheral bed.
C...
      DATA PRV/  1.33D+6,  9.99D99, 1.56D+8,  9.99D99,  9.99D99,
     *           1.54D+8,  3.83D+7, 1.52D+8,  9.99D99,  1.49D+8,
     *           9.99D99,  2.43D+8, 9.99D99,  7.04D+8,  9.99D99,
     *           9.99D99,  9.99D99, 1.46D+9,  9.99D99,  7.42D+7/
C...
C...  Initialize extramural pressure vector
C...
      DATA PEXT/ 0.0D+2,  0.0D+2,  0.0D+2,  0.0D+2,  0.0D+2,
     *           0.0D+2,  0.0D+2,  0.0D+2,  0.0D+2,  0.0D+2,
     *           0.0D+2,  0.0D+2,  0.0D+2,  0.0D+2,  0.0D+2,
     *           0.0D+2,  0.0D+2,  0.0D+0,  0.0D+0,  0.0D+0/
C...
C...  Set initial flows [m^3/sec]
C...
C...       Arterial flows
C...
      DATA   AQ/ 8.97D-5,  9.00D-5,  9.00D-5,  6.50D-5,  6.50D-5,
     *           6.50D-5,  5.67D-5,  2.33D-5,  1.50D-5,  1.50D-5,
     *           6.67D-6,  6.67D-6,  1.67D-6,  1.67D-6,  1.67D-6,
     *           1.67D-5,  1.67D-5,  1.67D-5,  1.58D-5,  1.58D-5/
C...
C...      Venous flows
C...
      DATA   VQ/ 9.00D-5,  2.00D-5,  7.50D-5,  7.00D-5,  7.00D-5,
     *           6.50D-5,  6.50D-5,  2.33D-5,  1.00D-5,  1.10D-5,
     *           4.70D-6,  7.50D-6,  4.50D-6,  1.67D-6,  1.67D-5,
     *           1.67D-5,  1.67D-5,  1.67D-5,  1.58D-5,  1.58D-5/
C...
C...     Venous output flows into heart
C...
      DATA  VOQ/ 9.0D-5, 1.7D-5, 7.3D-5/
C...
C...       Peripheral flows
C...
      DATA PQ /   8.97D-5,  0.00D00,  8.33D-6,  0.00D00,  0.00D00,
     *            8.33D-6,  3.33D-5,  8.33D-6,  0.00D00,  8.33D-6,
     *            0.00D00,  5.00D-6,  0.00D00,  1.67D-6,  0.00D00,
     *            0.00D00,  0.00D00,  8.33D-7,  0.00D00,  1.58D-5/
C...
C...  Initial Heart Flows
C...
      DATA HQ /   8.97D-5,  8.97D-5,  8.97D-5,  8.97D-5/
C...
C... The following arrays code the linkage between vascular segments
C... Each segment link element contains the index of the next segment
C... in the cardiovascular tree.  For example, ALIN(3) = 4, which
C... means arterial segment 3 feeds arterial segment 4. Segment -1 codes
C... for terminal peripheral beds for arteries and for the heart for veins.
C...
C...  SEGMENT   A1 A2 A3 A4 A5 A6 A7 A8 A9 A10 A11 A12 A13 A14
```

```
C...
      DATA ALIN/-1, 3, 4, 5, 6, 7, 8, 9,10, 11, 12, 13, 14, -1,
C...
C...  SEGMENT  A15 A16 A17 A18 A19 A20
C...
     *          16, 17, 18, 19, 20, -1/
C...
C... Venous linkage
C...
C...  SEGMENT    V1  V2 V3 V4 V5 V6 V7 V8  V9 V10 V11 V12 V13 V14
C...
      DATA VLIN/-1, -1, -1, 3, 4, 5, 6, 7,  8,  9, 10, 11, 12, 13,
C...
C...  SEGMENT   V15 V16 V17 V18 V19 V20
C...                                P20
     *            2, 15, 16, 17, 18, 19/
C...
C...  Peripheral Segment Linkage, PIN specifies the input source for the
C...  segment which is only from arteries, POUT specifies the output segment
C...  which is only to veins. "0" means there is no peripheral segment for the
C...  segment number.  Note that the pressure at the inlet of the peripheral
C...  segment is at AP(PIN(I) + 1)
C..
C...  SEGMENT      P1 P2 P3 P4 P5 P6 P7 P8  P9 P10 P11 P12 P13 P14
C...
      DATA PIN  / 0,  0, 3, 0, 0, 6, 7, 8,  0, 10,  0, 12,  0,  0,
C...
C...  SEGMENT   P15 P16 P17 P18 P19 P20
C...
     *            0,  0,  0, 18,  0,  0/
C..
C...  SEGMENT      P1  P2  P3 P4 P5 P6 P7 P8  P9 P10 P11 P12 P13 P14
C...
      DATA  POUT/ 1,  0,   3, 0, 0, 6, 7, 8,  0, 10,  0, 12,  0,  0,
C...
C...  SEGMENT   P15 P16 P17 P18 P19 P20
C...
     *            0,  0,  0, 18,  0,  0/
C...
C...  Z-axis positions relative to tricuspid valve for
C...  the origin of the arterial segment [m]
C...
      DATA ZAO/  0.0D0,   0.0D0,   5.D-2,   0.D-2,  -15.D-2,
     *         -20.D-2, -32.D-2, -42.D-2, -50.D-2,  -65.D-2,
     *         -85.D-2,-100.D-2,-125.D-2,-130.D-2,    5.D-2,
     *          15.D-2,  25.D-2,  34.D-2,  37.D-2,   41.D-2/
C...
C...  Z-axis positions relative to tricuspid valve for
C...  the termination of the arterial segment [m].
C...
      DATA ZAT/  3.D-2,  5.0D-2,   0.0D0, -15.D-2,  -20.D-2,
     *         -32.D-2, -42.D-2, -50.D-2, -65.D-2,  -85.D-2,
     *        -100.D-2,-125.D-2,-130.D-2,-132.D-2,   15.D-2,
     *          25.D-2,  34.D-2,  37.D-2,  41.D-2,   42.D-2/
C...
C...  Z-axis positions relative to tricuspid valve for
C...  the origin of the venous segment [m].
C...
      DATA ZVO/ 3.0D-2,1.5D-2, -1.5D-2, -15.D-2, -20.D-2,
     *        -32.D-2, -42.D-2, -50.D-2, -65.D-2, -85.D-2,
     *       -100.D-2,-125.D-2,-130.D-2,-132.D-2,  15.D-2,
     *         25.D-2,  34.D-2,  37.D-2,  41.D-2,  42.D-2/
C...
C...  Z-axis positions relative to tricuspid valve for
C...  the termination of the venous segment [m].
```

```
C...
       DATA ZVT/ 0.0D0,  0.0D00,  0.0D00, -1.5D-2,  -15.D-2,
     *          -20.D-2, -32.D-2, -42.D-2, -50.D-2,  -65.D-2,
     *          -85.D-2,-100.D-2,-125.D-2,-130.D-2,   1.5D-2,
     *           15.D-2,  25.D-2,  34.D-2,  37.D-2,   41.D-2/
C...
C... Initial orientations (degrees) of z-axis projection of vascular
C... segments.  Arterial and venous assumed to be at the same orientation.
C...
      DATA THETA/ 20.D0, 20.D0, 20.D0, 20.D0,  20.D0,  ! Partially reclined
     *            20.D0, 20.D0, 20.D0, 90.D0,  90.D0,  ! Seated position
     *            90.D0, 110.D0, 110.D0, 90.D0, 20.D0, ! 20 deg SBA
     *            0.D0, 0.D0, 0.D0, 0.D0,  0.D0/
C...
C... Define some physical constants
C...
      Pi     = DACOS(-1.0D0)          ! pi
      g0     = 9.80665D0              ! m/sec^2 - earth's gravity
      rho0   = 1050.0D0               ! kg/m^3  - density of whole blood
      mu0    = 2.70D-3                ! N-sec/m^2 - fluid viscosity
      D2R    = Pi/180.D0              ! scale from degrees to radians
      MM2PA  = 1.01325D5/760.D0       ! scale from mmHg to Pascals
      Gz     = 4.D0                   ! Initial Gz
      R2     = 0.002D0                ! Jaron's wall energy term
C...
C...  INITIAL CONDITIONS (T = 0)
C...
C...  The pressures are set assuming a prone posture ie transverse g0.
C...  The initial pressures are assigned by linearly interpolating
C...  pressures from the foot to the heart and from the heart to the
C...  cerebral segment. The mean arterial pressure is assumed
C...  to be 100 mmHg at the heart and 95 mm Hg at both the foot and
C...  cerebral segments.  The venous pressure is assumed to be 2 mmHg
C...  at the heart and 5 mmHg at both the foot and cerebral segments.
C...
      PAFOOT    =  95.D0*MM2PA
      PAHEAD    =  95.D0*MM2PA
      IFOOTSEG  = 14
      IHEADSEG  = 20
      IHEARTSEG =  1
      PRATRM =   PEXT(1)          ! Right Atrial Pressure
      HP(1)  = PRATRM + MM2PA     ! Inlet Pressure to Right Atrium
      HDP1   =   2.D0*MM2PA       ! Increase in right atrial pressure
      PRVENT =  HP(1) + HDP1      ! Inlet Pressure to Right Ventricular
      HP(2)  = PRVENT
      HDP2   = 11.D0*MM2PA        ! Increase in right ventricular pressure
      PLATRM =  4.D0*MM2PA       ! Left Atrial Pressure
      HP(3)  = PLATRM             ! Left Atrial Inlet Pressure
      HDP3   = 4.D0*MM2PA         !
      PLVENT = HP(3) + HDP3       ! Left Ventricular Pressure
      HP(4)  = PLVENT             ! Inlet Pressure to LV
      HDP4   = 92.D0*MM2PA        ! Increase in LV Pressure
C...
C...  G-Profile parameters
C...
      T0 = 0.D0
      GSTART = 1.D0
      GMAX = 3.D0
      TMAX = 400.D0
      TBRK1 = 10.D0
      TBRK2 = 300.D0
      GFIN =  3.D0
      i = 1
      DO WHILE (i .LE. NEQ)
         THETA(i) = THETA(i)*D2R
```

```
            i = i + 1
         END DO
         i = 1
         DO WHILE (i .LE. NEQ)
            SCALEF = 1.D0
            IF( i .LE. 5 .OR. i .EQ. 15 ) SCALEF = 33.D0
            ARES(i) = 81.D0*mu0*ALO(i)/(8.D0*Pi*ArU(i)**4)*SCALEF
            DPVS = DBLE(PVS(i))
            Vr(i) = DMAX1(Vr(i), 1.33D0*VrU(i))
            VRES(i) = 81.D0*mu0*(VLO(i)/DPVS)/(8.D0*Pi*Vr(i)**4)
            i = i + 1
         END DO
         AP(1) = HP(2) + HDP2 + 2.D0*MM2PA
         AP(2) = HP(4) + HDP4 + 2.D0*MM2PA
         i = 3
         DO WHILE ( i .LE. IFOOTSEG)
            AP(i) = AP(i-1) - AQ(i-1)*ARES(i-1)
     *              + rho0*Gz*g0*(ZAO(i)-ZAT(i))*DCOS(THETA(i))
            i = i + 1
         END DO
         AP(15) = AP(3)
         i = 16
         DO WHILE ( i .LE. IHEADSEG)
            AP(i) = AP(i-1) - AQ(i-1)*ARES(i-1)
     *              + rho0*Gz*g0*(ZAO(i)-ZAT(i))*DCOS(THETA(i))
            i = i + 1
         END DO
C...
C...  Set the initial radii and arterial capacitance
C...
C
         i = 1
         DO WHILE (i .LE. NEQ)
            dPwall  = AP(i)
            deltaR  = 7.5D-1*dPwall*ArU(i)**2/(AE(i)*Ah(i))
            Ar(i)   = ArU(i) + deltaR
            ACAP(i) = 3.D0*Pi*Ar(i)**3*ALO(i)/(2.D0*AE(i)*Ah(i))
            deltaR  = dPwall*ACAP(i)/(2.D0*Pi*Ar(i)*ALO(i))
            Ar(i)   = ArU(i) + deltaR
            AVOL(i) = Pi*Ar(i)**2*ALO(i)
            i = i + 1
         END DO
C...
C...  Initial Venous Pressures
C...
         VOP(2) = HP(1)
         VOP(3) = HP(1)
         VP(1) = HP(3)  + VQ(1)*VRES(1)
     *         - rho0*Gz*g0*(ZVO(1) - ZVT(1))*DCOS(THETA(1))
         VP(2) = VOP(2) + VQ(2)*VRES(2)
     *         - rho0*Gz*g0*(ZVO(2) - ZVT(2))*DCOS(THETA(2))
         VP(3) = VOP(3) + VQ(3)*VRES(3)
     *         - rho0*Gz*g0*(ZVO(3) - ZVT(3))*DCOS(THETA(3))
         i = 4
         DO WHILE ( i .LE. IFOOTSEG)
            VP(i) = VP(i-1) + VQ(i)*VRES(i)
     *              - rho0*Gz*g0*(ZVO(i) - ZVT(i))*DCOS(THETA(i))
            i = i + 1
         END DO
         VP(15) = VP(2) + VQ(15)*VRES(15)
     *                  - rho0*Gz*g0*(ZVO(15) - ZVT(15))*DCOS(THETA(15))
         i = 16
         DO WHILE ( i .LE. IHEADSEG)
            VP(i) = VP(i-1) + VQ(i)*VRES(i)
     *              - rho0*Gz*g0*(ZVO(i) - ZVT(i))*DCOS(THETA(i))
```

```
         i = i + 1
      END DO
C...
C...  Compute the initial resistances
C...
      i = 1
      DO WHILE (i .LE. NEQ)
         VCAP(i) = VCAPO(i)
         dPwall  = VP(i) - PEXT(i)
         deltaR  = dPwall*VCAP(i)/(2.DO*Pi*VrU(i)*VLO(i))
         Vr(i)   = VrU(i) + deltaR
         deltaR  = dPwall*VCAP(i)/(2.DO*Pi*Vr(i)*VLO(i))
         Vr(i)   = VrU(i) + deltaR
         VVOL(i) = Pi*Vr(i)**2*VLO(i)
         i = i + 1
      END DO

C...
C...   Initial Heart flows
C...
      HQ(1) = 9.D-5
      HQ(2) = 9.D-5
      HQ(3) = 9.D-5
      HQ(4) = 9.D-5

C...
C...  Set initial Peripheral Pressures
C...
      PP(1) =  PQ(1)*PRA(1) + VQ(1)*PRV(1) + VP(1)
      i = 3
      DO WHILE ( i .LE. NEQ)
         IF( PIN(i) .GT. 0 )  THEN
             PP(i) = AP(i+1)
         END IF
         i = i + 1
      END DO
      PP(14) =  PQ(14)*PRA(14) + VQ(14)*PRV(14) + VP(14)
      PP(20) =  PQ(20)*PRA(20) + VQ(20)*PRV(20) + VP(20)
C...
C...   Call DERV to set initial derivatives -- loop to stabilize derivatives
C...
      i = 1
      DO WHILE (i .LE. 100)
         CALL DERV
         i = i + 1
      END DO
      IP=0
      RETURN
      END
C...
      SUBROUTINE DERV
C...
C...  DERV CALCULATES THE TIME DERIVATIVES TO BE INTEGRATED BY RKF45
C...
C...  ODE COMMON
C...
C...   /Y/ time variables
C...   /F/ time derivatives of variables
C...   /S/ spatial derivatives of variables
C...   /R/ & /I/ real and integer parameters required to define constants and
C...             define the spatial integration grid.
C...
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      PARAMETER (NEQ = 20, NPSEG = 10)
      INTEGER NSTOP, NORUN, IP
```

Program CVSUBS.FOR

```
      INTEGER*2 ALIN, ALOUT, VLIN, VLOUT, PVS, PIN, POUT
      DOUBLE PRECISION MM2PA, mu0
      COMMON/T/          T,     NSTOP,     NORUN    ! Run Parameters
C...
C...  Arrays for segmental variables Pressure (P), Flow(Q), and radii (r)
C...
     1 /Y/    HP(4),      HQ(4),                    ! Heart's Chambers RA=1
     *      AP(NEQ),   AQ(NEQ),    Ar(NEQ),         ! Arterial P, Q, r
     *      VP(NEQ),   VQ(NEQ),    Vr(NEQ),         ! Venous P, Q, r
     *      VOP(3),     VOQ(3),                     ! Venous flows into heart
     *      PP(NEQ),   PQ(NEQ),                     ! Peripheral P, Qin, Qout
C...
C... Time derivatives of the segmental variables: Pt, Qt, rt
C...
     2 /F/   HPt(4),     HQt(4),                    ! Heart's Chambers RA=1
     *     APt(NEQ),  AQt(NEQ),   Art(NEQ),         ! Arterial Pt, Qt, rt
     *      VPt(NEQ),  VQt(NEQ),   Vrt(NEQ),        ! Venous Pt, Qt, rt
     *     VOPt(3),    VOQt(3),                     ! Venous flows into heart
     *     PPt(NEQ),  PQt(NEQ),                     ! Peripheral Pt, QINt, QOUTt
C...
C... Parameters necessary to form the differential equations
C...
C...    Pi -  3.14159...
C...    g0 -  9.80665 [m/sec^2]  earth's acceleration of gravity
C...  rho0 -  1050.   [kg/m^3]   density of whole blood (45% Hct)  | Assumed
C...   mu0 -  2.7     [cp]       viscosity of whole blood (45% Hct)| Constant
C...   D2R -  pi/180  [radians/degree] scale factor
C...
     3 /R/  Pi, g0, rho0, mu0,  D2R, MM2PA, R2,      ! Constants
     *       ZAO(NEQ), ZAT(NEQ), ZVO(NEQ), ZVT(NEQ), ! Arterial & Venous
     *       THETA(NEQ), HCAP(4), HVOL(4), HVR(4),                    ! Orientation angle
     *       ALO(NEQ), ArU(NEQ),  AE(NEQ), Ah(NEQ),   ! Arterial l, r, E, h
     *       ACAP(NEQ), ARES(NEQ), AINERT(NEQ),       ! Arterial Capacitance, resistance
     *       VLO(NEQ),  VrO(NEQ),  VrU(NEQ),          ! Venous l, r, rUNSTRESSED
     *       VCAPO(NEQ), VCAP(NEQ), VRES(NEQ), VINERT(NEQ), ! Venous Capacitance, resistance, inertance
     *       PRA(NEQ), PRV(NEQ), PCAP(NEQ),           ! Peripheral Ra, Rv, C
     *       PINERT(NEQ), PVOL(NEQ),                  ! Peripheral I, V
     *       PAO(NEQ),  PVO(NEQ),                     ! Initial P conditions
     *       QAO(NEQ),  QVO(NEQ),                     ! Initial Q conditions
     *       AVOL(NEQ),  VVOL(NEQ),                   ! A & V Volumes
     *       PEXT(NEQ),                               ! Externally applied Pressure
     *       TO, GSTART, GMAX, TBRK1, TBRK2, TMAX, GFIN, ! G-Profile parameters
     *       Gz, ADPG(NEQ), VDPG(NEQ)                 ! Gz & Delta P from G
     4 /I/  IP, NDXPER(NPSEG),                        ! Peripheral Bed Indexes
     *       ALIN(NEQ), ALOUT(NEQ),                   ! Linkage Data arterial
     *       VLIN(NEQ), VLOUT(NEQ),                   ! Venous
     *       PIN(NEQ), POUT(NEQ),                     ! Peripheral
     *       PVS(NEQ),                                ! Number parallel venous segments
     *       IFOOTSEG, IHEADSEG, IHEARTSEG            ! Foot, Head, & Heart seg nums
      DIMENSION PQOUT(NEQ)
C...
C...   Right Heart
C...
      PRATRM =    PEXT(1)        ! Right Atrial Pressure
      HP(1)  = PRATRM + MM2PA    ! Inlet Pressure to Right Atrium
      HDP1   =    1.D0*MM2PA     ! Increase in right atrial pressure
      PRVENT =  HP(1) + HDP1     ! Inlet Pressure to Right Ventricular
      HP(2)  = PRVENT
      HDP2   = 11.D0*MM2PA       ! Increase in right ventricular pressure
      PINTHO =  -1.0D0 *MM2PA    ! Intra-thoracic Pressure                    ! Extramural
pressure for the pulmonary bed
C...
C...  Left Heart
C...
      PLATRM =   4.D0*MM2PA      ! Left Atrial Pressure
```

```
      HP(3)  = PLATRM            ! Left Atrial Inlet Pressure
      HDP3   = 4.D0*MM2PA        !
      PLVENT = HP(3) + HDP3      ! Left Ventricular Pressure
      HP(4)  = PLVENT            ! Inlet Pressure to LV
      HDP4   = 92.D0*MM2PA       ! Increase in LV Pressure
C...
C...   Ensure pressures are not less than external pressures
C...
      i = 1
      DO WHILE ( i .LE. NEQ )
         IF (VQ(i) .LT. 0.D0) THEN
             VQ(i)  = 0.D0 ! Venous valves
         END IF
         i = i + 1
      END DO
      i = 1
      DO WHILE (i .LE. 3)
         VOQ(i) = DMAX1(VOQ(i), 0.D0)
         i = i + 1
      END DO
      AQ(1) = DMAX1(AQ(1), 0.D0) ! No reverse flow into the heart
      AQ(2) = DMAX1(AQ(2), 0.D0) ! from arteries or out of the heart
      HQ(1) = DMAX1(HQ(1), 0.D0) ! into veins
      HQ(3) = DMAX1(HQ(3), 0.D0)
C...
C...  Set the Heart level G level
C...
C     CALL TRAPG(T0,GSTART,GMAX,TMAX,TBRK1,TBRK2,GFIN,T,Gz)
C...
C...  Set peripheral inlet pressures to the appropriate
C...  arterial outlet pressures
C...
      i = 3
      DO WHILE ( i .LE. NEQ)
         IF( PIN(i) .GT. 0 )  THEN
             PP(i) = AP(i+1)
             PPt(i) = APt(i+1)
         END IF
         i = i + 1
      END DO
C...
C...  Compute the resistance, capacitance and inertance for the arterial segments
C...
      AP(15) = AP(3)
      i = 1
      DO WHILE (i .LE. NEQ)
         IF ( AP(i) .LE. PEXT(i) ) THEN
            Ar(i) = ArU(i)   ! If arterial pressure is below external
         END IF      ! pressure set radius to minimum (unstressed).
C...
C...   rationalize pressure, radius, and capacitance.
C...
         dPwall = AP(i) - PEXT(i)
         IF(dPwall .LE. 0.D0) Ar(i) = ArU(i)
         ACAP(i) = 3.D0*Pi*Ar(i)**3*ALO(i)/(2.D0*AE(i)*Ah(i))
         deltaR  = dPwall*ACAP(i)/(2.D0*Pi*Ar(i)*ALO(i))
C         Ar(i)   = ArU(i) + deltaR
         AVOL(i) = Pi*Ar(i)**2*ALO(i)
         SCALEF = 1.D0
         IF( i .LE. 5 .OR. i .EQ. 15 ) SCALEF = 33.D0
         ARES(i)   = 81.D0*muO*ALO(i)/(8.D0*Pi*Ar(i)**4)*SCALEF
         AINERT(i) =  9.D0*rho0*ALO(i)**2/(4 D0*AVOL(i))
         ADPG(i)   = rho0*Gz*g0*(ZAO(i) - ZAT(i))*DCOS(THETA(i))
         i = i + 1
      END DO
```

Program CVSUBS.FOR

```
C...
C...  Venous resistance, capacitance and inertance for the venous segments
C...
      i = 1
      DO WHILE (i .LE. NEQ)
         dPwall = DMAX1(VP(i) - PEXT(i),0.D0)
         IF(dPwall .LE. 0.D0) Vr(i) = VrU(i)
         Vrmax = 1.333*VrU(i)
         Vr(i)  = DMAX1(Vr(i), VrU(i))
         Vr(i)  = DMIN1(Vr(i), Vrmax)
         DPVS = DBLE(PVS(i))
         VVOL(i) = Pi*Vr(i)**2*VLO(i)
         VRES(i) = 81.D0*mu0*(VLO(i)/DPVS)/(8.D0*Pi*Vr(i)**4)
         VINERT(i) = 9.D0*rho0*VLO(i)**2/(4.D0*VVOL(i))
         VDPG(i) = rho0*Gz*g0*(ZVO(i) - ZVT(i))*DCOS(THETA(i))
         i = i + 1
      END DO
      CAPL = 1.D-2 ! Capillary Length
      i = 1
      DO WHILE( i .LE. NPSEG)
         j = NDXPER(i)
         dPwall = DMAX1(PP(j) - PEXT(j), MM2PA)  ! Min pressure 1 mmHg
         PVOL(j) = PCAP(j) * dPwall
         i = i + 1
      END DO
C...
C...    Define the differential equations describing pressure, flow, and
C...    radius
C...
C...    Pt(t) = 1/C*(Qin(t) - Qout(t))  + R2/C*(Qtin(t)- Qtout(t))
C...    Qt(t) = 1/L*(Pin(t) - Pout(t) + PGz - Pext - R*Q(t))
C...    rt(t) = 1/(2*Pi*r*l)*(Qin(t) - Qout(t))
C...
C...
C...      Infow to the right atrium HQ(1) = VOQ(2) + VOQ(3)
C...
        IF (VQ(2) .LE. 0.D0) VP(2) = HP(1)
        VOQt(2) = 1.D0/VINERT(2)*(VP(2) - HP(1) + VDPG(2)   ! Outflow from the Superior V.C.
     *          - VRES(2)*VOQ(2))
        IF (VQ(3) .LE. 0.D0) VP(3) = HP(1)
        VOQt(3) = 1.D0/VINERT(3)*(VP(3) - HP(1) + VDPG(3)   ! Outflow from the Inferior V.C.
     *          - VRES(3)*VOQ(3))
        HQt(1) = VOQt(2) + VOQt(3)
        HQ(1) = VOQ(2) + VOQ(3)

C...
C...   Right Atrium   [Heart Segment 1]
C...
      HQ2 = HQ(1) - HQ(2)
      HQ2t = HQt(1) - HQt(2)
      HR2 = 2.D-5/HCAP(1)
      HPt(1) = HQ2t*HR2 + HQ2/HCAP(1)
      HPt(1) = 0.D0
      HPt(2) = 0.D0
      HQ(2) = (HP(1) + HDP1 + MM2PA - HP(2))/HVR(1)
      HQt(2) = (HPt(1) - HPt(2))/HVR(1)
      HQt(2) = HQt(1)
      HQ(2) =  HQ(1)
C...
C...  Right Ventricle   [Heart Segment 2]
C...
      HQ2 = HQ(2) - AQ(1)
      HQ2t = HQt(2) - AQt(1)
      HR2 = 2.D-5/HCAP(2)
      HPt(2) = HQ2t*HR2 + HQ2/HCAP(2)
```

Program CVSUBS.FOR

```
      HPt(2) = 0.D0
C...
C...   Pulmonary Circulation
C...
C...  Pulmonary Artery    [Arterial Segment 1]
C...
      AQ2  = AQ(1) - PQ(1)
      AQ2t = AQt(1) - PQt(1)
      AR2 = R2/ACAP(1)
      APt(1) = AQ2t*AR2 + AQ2/ACAP(1)
      APt(1) = 0.D0
      Art(1) =  1.D0/(2.D0*Pi*Ar(1)*ALO(1))*AQ2
      Art(1) =  0.D0
      AQ(1) = (HP(2) + HDP2 + 2.D0*MM2PA - AP(1))/HVR(2)
      AQt(1) = (HPt(2) - APt(1))/HVR(2)
      AQ(1) = HQ(2)
C...
C...  Pulmonary Capillary Bed   [Peripheral Segment 1]
C...
      PQt(1) = 1.D0/AINERT(1)*(AP(1) - PP(1) + ADPG(1)
     *          - ARES(1)*PQ(1))
      PQ2 = PQ(1) - VQ(1)
      PPt(1) = PRA(1)*PQt(1) + PQ2/PCAP(1)
C...
C...  Pulmonary Veins
C...
      VQt(1) = (PPt(1) - VPt(1) - PQt(1)*PRA(1))/PRV(1)
      VQ2  = VQ(1)  - HQ(3)
      VQ2t = VQt(1) - HQt(3)
      VR2 = R2/VCAP(1)
      VPt(1) = VQ2t*VR2 + VQ2/VCAP(1)
      Vrt(1) =  1.D0/(2.D0*Pi*Vr(1)*VLO(1))*VQ2
C      Vrt(1) =  0.D0
C...
C...  Left Atrium   [Heart Segment 3]
C...
      HQt(3) = 1.D0/VINERT(1)*(VP(1) - HP(3) + VDPG(1)
     *          - HQ(3)*VRES(1))
      VOQt(1) = HQt(3)
      HQ2 = HQ(3) - HQ(4)
      HQ2t = HQt(3) - HQt(4)
      HR2 = 2.D-5/HCAP(3)
      HPt(3) = HQ2t*HR2 + HQ2/HCAP(3)
      HPt(3) = 0.D0
C...
C...  Left Ventricle   [Heart Segment 4]
C...
      HQ(4) = (HP(3) + HDP3 + 2.D0*MM2PA - HP(4))/HVR(3)
      HQ(4) = HQ(3)
      HQt(4) = (HPt(3) - HPt(4))/HVR(3)
      HQt(4) = HQt(3)
      HQ2 = HQ(4) - AQ(2)
      HQ2t = HQt(4) - AQt(2)
      HR2 = 2.D-5/ACAP(4)
      HPt(4) = HQ2t*HR2 + HQ2/HCAP(4)
      HPt(4) = 0.D0
C...
C...  Ascending Aortic artery [Arterial Segment 2]
C...
C...  AQ(2) = (HP(4) + HDP4 + 2.D0*MM2PA - AP(2))/HVR(4)
      AQt(2) = (HPt(4) - APt(2))/HVR(4)
      AQ2 = AQ(2) - AQ(3) - AQ(15)
      AQ2t = AQt(2) - AQt(3) - AQt(15)
      AR2 = R2/ACAP(2)
      APt(2) = AQ2t*AR2 + AQ2/ACAP(2)
```

```
      APt(2) = 0.D0
      Art(2) = 1.D0/(2.D0*Pi*Ar(2)*ALO(2))*AQ2
      Art(2) = 0.D0
C...
C...  Descending Aorta   [Arterial Segment 3]
C...
      AQt(3) =  1.D0/AINERT(2)*(AP(2) - AP(3)
     *          + ADPG(2) - ARES(2)*(AQ(3) + AQ(15))) - AQt(15)
      AQ2  =  AQ(3) - AQ(4)  - PQ(3)
      AQ2t = AQt(3) - AQt(4) - PQt(3)
      AR2 = R2/ACAP(3)
      APt(3) = AQ2t*AR2 + AQ2/ACAP(3)
      AQt(4) = 1.D0/AINERT(3)*(AP(3) - AP(4) + ADPG(3)
     *          - ARES(3)*(AQ(4) + PQ(3))) - PQt(3)
      Art(3) =  1.D0/(2.D0*Pi*Ar(3)*ALO(3))*AQ2
      Art(3) = 0.D0
C...
C...  Form the derivatives for the other arterial segments below the heart.
C...
C...  Thoracic and Cardiac   [Arterial Segment 4]
C...
      AQ2 = AQ(4) - AQ(5)
      AQ2t = AQt(4) - AQt(5)
      AR2 = R2/ACAP(4)
      APt(4) =  1.D0/ACAP(4)*AQ2 + AR2*AQ2t
      AQt(5) =  1.D0/AINERT(4)*(AP(4) - AP(5)
     *       + ADPG(4) - ARES(4)*AQ(5))
      Art(4) =  1.D0/(2.D0*Pi*Ar(4)*ALO(4))*AQ2
      Art(4) = 0.D0
C...
C...  Diaphragm  [Arterial Segment 5]
C...
      AQ2 = AQ(5) - AQ(6)
      AQ2t = AQt(5) - AQt(6)
      AR2 = R2/ACAP(5)
      APt(5) =  AR2*AQ2t + AQ2/ACAP(5)
      AQt(6) =  1.D0/AINERT(5)*(AP(5) - AP(6)
     *       + ADPG(5) - ARES(5)*AQ(6))
      Art(5) =  1.D0/(2.D0*Pi*Ar(5)*ALO(5))*AQ2
      Art(5) = 0.D0
C...
C...  Renal - Hepatic   [Arterial Segment 6]
C...
      AQ2 = AQ(6) - AQ(7) - PQ(6)
      AQ2t = AQt(6) - AQt(7) - PQt(6)
      AR2 = R2/ACAP(6)
      APt(6) =  AR2*AQ2t + AQ2/ACAP(6)
      AQt(7) =  1.D0/AINERT(6)*(AP(6) - AP(7)
     *       + ADPG(6) - ARES(6)*(AQ(7) + PQ(6))) - PQt(6)
      Art(6) =  1.D0/(2.D0*Pi*Ar(6)*ALO(6))*AQ2
      Art(6) = 0.D0
C...
C...  Splanchnic  [Arterial Segment 7]
C...
      AQ2 = AQ(7) - AQ(8) - PQ(7)
      AQ2t = AQt(7) - AQt(8) - PQt(7)
      AR2 = R2/ACAP(7)
      APt(7) =  AR2*AQ2t + AQ2/ACAP(7)
      AQt(8) =  1.D0/AINERT(7)*(AP(7) - AP(8)
     *       + ADPG(7) - ARES(7)*(AQ(8) + PQ(7))) - PQt(7)
      Art(7) =  1.D0/(2.D0*Pi*Ar(7)*ALO(7))*AQ2
      Art(7) = 0.D0
C...
C...  Buttocks   [Arterial Segment 8]
C...
```

```
      AQ2 = AQ(8) - AQ(9) - PQ(8)
      AQ2t = AQt(8) - AQt(9) - PQt(8)
      AR2 = R2/ACAP(8)
      APt(8) =  1.D0/ACAP(8)*AQ2 + AR2*AQ2t
      AQt(9) =  1.D0/AINERT(8)*(AP(8) - AP(9)
     *        + ADPG(8) - ARES(8)*(AQ(9) + PQ(8))) - PQt(8)
      Art(8) =  1.D0/(2.D0*Pi*Ar(8)*ALO(8))*AQ2
      Art(8) = 0.D0
C...
C...  Femoralis   [Arterial Segment 9]
C...
      AQ2 = AQ(9) - AQ(10)
      AQ2t = AQt(9) - AQt(10)
      AR2 = R2/ACAP(9)
      APt(9) =  AR2*AQ2t + AQ2/ACAP(9)
      AQt(10) =  1.D0/AINERT(9)*(AP(9) - AP(10)
     *        + ADPG(9) - ARES(9)*AQ(10))
      Art(9) =  1.D0/(2.D0*Pi*Ar(9)*ALO(9))*AQ2
      Art(9) = 0.D0
C...
C...  Thigh   [Arterial Segment 10]
C...
      AQ2 = AQ(10) - AQ(11) - PQ(10)
      AQ2t = AQt(10) - AQt(11) - PQt(10)
      AR2 = R2/ACAP(10)
      APt(10) =  AR2*AQ2t + AQ2/ACAP(10)
      AQt(11) =  1.D0/AINERT(10)*(AP(10) - AP(11)
     *        + ADPG(10) - ARES(10)*(AQ(11)+PQ(10))) - PQt(10)
      Art(10) =  1.D0/(2.D0*Pi*Ar(10)*ALO(10))*AQ2
      Art(10) = 0.D0
C...
C...  Knee   [Arterial Segment 11]
C...
      AQ2 = AQ(11) - AQ(12)
      AQ2t = AQt(11) - AQt(12)
      AR2 = R2/ACAP(11)
      APt(11) =  AQ2/ACAP(11) +  AR2*AQ2t
      AQt(12) =  1.D0/AINERT(11)*(AP(11) - AP(12)
     *              + ADPG(11) - ARES(11)*AQ(12))
      Art(11) =  1.D0/(2.D0*Pi*Ar(11)*ALO(11))*AQ2
      Art(11) =  0.D0
C...
C...  Calf   [Arterial Segment 12]
C...
      AQ2 = AQ(12) - AQ(13) - PQ(12)
      AQ2t = AQt(12) - AQt(13) - PQt(12)
      AR2 = R2/ACAP(12)
      APt(12) =  AR2*AQ2t + AQ2/ACAP(12)
      AQt(13) =  1.D0/AINERT(12)*(AP(12) - AP(13)
     *        + ADPG(12) - ARES(12)*(AQ(13)+PQ(12))) - PQt(12)
      Art(12) =  1.D0/(2.D0*Pi*Ar(12)*ALO(12))*AQ2
      Art(12) = 0.D0
C...
C...  Ankle   [Arterial Segment 13]
C...
      AQ2 = AQ(13) - AQ(14)
      AQ2t = AQt(13) - AQt(14)
      AR2 = R2/ACAP(13)
      APt(13) =  AQ2t*AR2 + AQ2/ACAP(13)
      AQt(14) =  1.D0/AINERT(13)*(AP(13) - AP(14)
     *        + ADPG(13) - ARES(13)*AQ(14))
      Art(13) =  1.D0/(2.D0*Pi*Ar(13)*ALO(13))*AQ2
      Art(13) = 0.D0
C...
C...   Foot   [Arterial Segment 14]
```

```
C...
      AQ2 = AQ(14) - PQ(14)
      AQ2t = AQt(14) - PQt(14)
      AR2 = R2/ACAP(14)
      APt(14) = AR2*AQ2t + AQ2/ACAP(14)
      PQt(14) = 1.D0/AINERT(14)*(AP(14) - PP(14) + ADPG(14)
     *          - ARES(14)*PQ(14))
      Art(14) =  1.D0/(2.D0*Pi*Ar(14)*ALO(14))*AQ2
      Art(14) = 0.D0
C...
C...   Peripheral Bed in Foot   [Peripheral Segment 14]
C...
      VQ(14) = (PP(14) - VP(14) - PQ(14)*PRA(14))/PRV(14)
      PQ2 = PQ(14) - VQ(14)
      PPt(14) = PQt(14)*PRA(14) + PQ2/PCAP(14)
C...
C...  Venous Drainage from Foot   [Venous Segment 14]
C...
      VQt(14) = (PPt(14) - VPt(14) - PQt(14)*PRA(14))/PRV(14)
      VQ2 = VQ(14) - VQ(13)
      VQ2t = VQt(14) - VQt(13)
      VR2 = R2/VCAP(14)
      VPt(14) = VQ2t*VR2 + VQ2/VCAP(14)
      VQt(13) = 1.D0/VINERT(14)*(VP(14) - VP(13) + VDPG(14)
     *          - VRES(14)*VQ(13))
      Vrt(14) =  1.D0/(2.D0*Pi*Vr(14)*VLO(14))*VQ2
C...
C...  Ankle  [Venous Segment 13]
C...
      PQOUT(12) = (PP(12) - VP(12) - PRA(12)*PQ(12))/PRV(12)
      PQOUTt = (PPt(12) - VPt(12) - PRA(12)*PQt(12))/PRV(12)
      VQ2 =  VQ(13) - (VQ(12) - PQOUT(12))
      VQ2t = VQt(13) - (VQt(12) - PQOUTt)
      VR2  = R2/VCAP(13)
      VPt(13) = VQ2t*VR2 + VQ2/VCAP(13)
      VQt(12) = 1.D0/VINERT(13)*(VP(13) - VP(12) + VDPG(13)
     *          - VRES(13)*(VQ(12) - PQOUT(12))) + PQOUTt
      Vrt(13) =  1.D0/(2.D0*Pi*Vr(13)*VLO(13))*VQ2
C...
C...  Calf   [Venous Segment 12]
C...
      VQ2 = VQ(12) - VQ(11)
      VQ2t = VQt(12) - VQt(11)
      VR2 = R2/VCAP(12)
      VPt(12) = VQ2t*VR2 + VQ2/VCAP(12)
      VQt(11) = 1.D0/VINERT(12)*(VP(12) - VP(11) + VDPG(12)
     *          - VRES(12)*VQ(11))
      Vrt(12) =  1.D0/(2.D0*Pi*Vr(12)*VLO(12))*VQ2
C...
C...  Knee   [Venous Segment 11]
C...
      PQOUT(10) = (PP(10) - VP(10) - PRA(10)*PQ(10))/PRV(10)
      PQOUTt = (PPt(10) - VPt(10) - PRA(10)*PQt(10))/PRV(10)
      VQ2 =  VQ(11) - (VQ(10) - PQOUT(10))
      VQ2t = VQt(11) - (VQt(10) - PQOUTt)
      VR2  = R2/VCAP(11)
      VPt(11) = VQ2t*VR2 + VQ2/VCAP(11)
      VQt(10) = 1.D0/VINERT(11)*(VP(11) - VP(10) + VDPG(11)
     *          - VRES(11)*(VQ(10) - PQOUT(10))) + PQOUTt
      Vrt(11) =  1.D0/(2.D0*Pi*Vr(11)*VLO(11))*VQ2
C...
C... .Thigh   [Venous Segment 10]
C...
      VQ2 = VQ(10) - VQ(9)
      VQ2t = VQt(10) - VQt(9)
```

```
      VR2 = R2/VCAP(10)
      VPt(10) = VQ2t*VR2 + VQ2/VCAP(10)
      VQt(9) = 1.D0/VINERT(10)*(VP(10) - VP(9) + VDPG(10)
     *         - VRES(10)*VQ(9))
      Vrt(10) =  1.D0/(2.D0*Pi*Vr(10)*VLO(10))*VQ2
C...
C...   Femoralis  [Venous Segment 9]
C...
      PQOUT(8) = (PP(8) - VP(8) - PRA(8)*PQ(8))/PRV(8)
      PQOUTt = (PPt(8) - VPt(8) - PRA(8)*PQt(8))/PRV(8)
      VQ2 =  VQ(9) - (VQ(8) - PQOUT(8))
      VQ2t = VQt(9) - (VQt(8) - PQOUTt)
      VR2  = R2/VCAP(9)
      VPt(9) = VQ2t*VR2 + VQ2/VCAP(9)
      VQt(8) = 1.D0/VINERT(9)*(VP(9) - VP(8) + VDPG(9)
     *         - VRES(9)*(VQ(8) - PQOUT(8))) + PQOUTt
      Vrt(9) =  1.D0/(2.D0*Pi*Vr(9)*VLO(9))*VQ2
C...
C...   Buttocks   [Venous Segment 8]
C...
      PQOUT(7) = (PP(7) - VP(7) - PRA(7)*PQ(7))/PRV(7)
      PQOUTt = (PPt(7) - VPt(7) - PRA(7)*PQt(7))/PRV(7)
      VQ2 =  VQ(8) - (VQ(7) - PQOUT(7))
      VQ2t = VQt(8) - (VQt(7) - PQOUTt)
      VR2  = R2/VCAP(8)
      VPt(8) = VQ2t*VR2 + VQ2/VCAP(8)
      VQt(7) = 1.D0/VINERT(8)*(VP(8) - VP(7) + VDPG(8)
     *         - VRES(8)*(VQ(7) - PQOUT(7))) + PQOUTt
      Vrt(8) =  1.D0/(2.D0*Pi*Vr(8)*VLO(8))*VQ2
C...
C...   Splanchnic   [Venous Segment 7]
C...
      PQOUT(6) = (PP(6) - VP(6) - PRA(6)*PQ(6))/PRV(6)
      PQOUTt = (PPt(6) - VPt(6) - PRA(6)*PQt(6))/PRV(6)
      VQ2 =  VQ(7) - (VQ(6) - PQOUT(6))
      VQ2t = VQt(7) - (VQt(6) - PQOUTt)
      VR2  = R2/VCAP(7)
      VPt(7) = VQ2t*VR2 + VQ2/VCAP(7)
      VQt(6) = 1.D0/VINERT(7)*(VP(7) - VP(6) + VDPG(7)
     *         - VRES(7)*(VQ(6) - PQOUT(6))) + PQOUTt
      Vrt(7) =  1.D0/(2.D0*Pi*Vr(7)*VLO(7))*VQ2
C...
C...   Renal - Hepatic   [Venous Segment 6]
C...
      VQ2 = VQ(6) - VQ(5)
      VQ2t = VQt(6) - VQt(5)
      VR2 = R2/VCAP(6)
      VPt(6) = VQ2t*VR2 + VQ2/VCAP(6)
      VQt(5) = 1.D0/VINERT(6)*(VP(6) - VP(5) + VDPG(6)
     *         - VRES(6)*VQ(5))
      Vrt(6) =  1.D0/(2.D0*Pi*Vr(6)*VLO(6))*VQ2
C...
C...   Diaphragm   [Venous Segment 5]
C...
      VQ2 = VQ(5) - VQ(4)
      VQ2t = VQt(5) - VQt(4)
      VR2 = R2/VCAP(5)
      VPt(5) = VQ2t*VR2 + VQ2/VCAP(5)
      VQt(4) = 1.D0/VINERT(5)*(VP(5) - VP(4) + VDPG(5)
     *         - VRES(5)*VQ(4))
      Vrt(5) =  1.D0/(2.D0*Pi*Vr(5)*VLO(5))*VQ2
C...
C...   Thoracic Circulation [Venous Segment 4]
C...
        PQOUT(3) = (PP(3) - VP(3) - PRA(3)*PQ(3))/PRV(3)
```

```
         PQOUTt = (PPt(3) - VPt(3) - PRA(3)*PQt(3))/PRV(3)
         VQ2 =  VQ(4) - (VQ(3) - PQOUT(3))
         VQ2t = VQt(4) - (VQt(3) - PQOUTt)
         VR2  = R2/VCAP(4)
         VPt(4) = VQ2t*VR2 + VQ2/VCAP(4)
         VQt(3) = 1.D0/VINERT(4)*(VP(4) - VP(3) + VDPG(4)
     *            - VRES(4)*(VQ(3) - PQOUT(3))) + PQOUTt
         Vrt(4) =  1.D0/(2.D0*Pi*Vr(4)*VLO(4))*VQ2
C...
C...     Inferior Vena Cava   [Venous Segment 3]
C...
         VQ2  = VQ(3)  - VOQ(3)
         VQ2t = VQt(3) - VOQt(3)
         VR2  = R2/VCAP(3)
         VPt(3) = VQ2t*VR2 + VQ2/VCAP(3)
         Vrt(3) = 1.D0/(2.D0*Pi*Vr(3)*VLO(3))*VQ2
C...
C...
C...    Above the heart
C...
C...     Subclavian - Upper Thorax   [Arterial Segment 15]
C...
        AP(15) = AP(3)
        AQ2 = AQ(15) - AQ(16)
        AQ2t = AQt(15) - AQt(16)
        AR2 = R2/ACAP(15)
        APt(15) = AQ2t*AR2 + AQ2/ACAP(15)
        AQt(15) =  1.D0/AINERT(2)*(AP(2) - AP(15)
     *           +  ADPG(2) - ARES(2)*(AQ(3) + AQ(15))) - AQt(3)
        AQt(16) = 1.D0/AINERT(15)*(AP(15) - AP(16) + ADPG(15)
     *           - ARES(15)*AQ(16))
        Art(15) =  1.D0/(2.D0*Pi*Ar(15)*ALO(15))*AQ2
        Art(15) = 0.D0
C...
C...    Lower Neck   [Arterial Segment 16]
C...
        AQ2 = AQ(16) - AQ(17)
        AQ2t = AQt(16) - AQt(17)
        AR2 = R2/ACAP(16)
        APt(16) = AQ2t*AR2 + AQ2/ACAP(16)
        AQt(17) =  1.D0/AINERT(16)*(AP(16) - AP(17)
     *       +  ADPG(16) - ARES(16)*AQ(17))
        Art(16) =  1.D0/(2.D0*Pi*Ar(16)*ALO(16))*AQ2
         Art(16) = 0.D0
C...
C...      Upper Neck (Carotid sinus)   [Arterial Segment 17]
C...
        AQ2 =  AQ(17) - AQ(18)
        AQ2t = AQt(17) - AQt(18)
        AR2 = R2/ACAP(17)
        APt(17) = AQ2t*AR2 + AQ2/ACAP(17)
        AQt(18) = 1.D0/AINERT(17)*(AP(17) - AP(18)
     *          + ADPG(17) - ARES(17)*AQ(18))
        Art(17) =  1.D0/(2.D0*Pi*Ar(17)*ALO(17))*AQ2
        Art(17) = 0.D0
C...
C...     Opthalmic   [Arterial Segment 18]
C...
        AQ2 = AQ(18) - (AQ(19) + PQ(18))
        AQ2t = AQt(18) - (AQt(19) + PQt(18))
        AR2 = R2/ACAP(18)
        APt(18) = AQ2t*AR2 + AQ2/ACAP(18)
        AQt(19) = 1.D0/AINERT(18)*(AP(18) - AP(19) + ADPG(18)
     *           - ARES(18)*(AQ(19) + PQ(18))) - PQt(18)
        Art(18) =  1.D0/(2.D0*Pi*Ar(18)*ALO(18))*AQ2
```

Program CVSUBS.FOR

```
      Art(18) =  0.D0
C...
C...  Midbrain   [Arterial Segment 19]
C...
      AQ2 = AQ(19) - AQ(20)
      AQ2t = AQt(19) - AQt(20)
      AR2 = R2/ACAP(19)
      APt(19) = AQ2t*AR2 + AQ2/ACAP(19)
      AQt(20) = 1.D0/AINERT(19)*(AP(19) - AP(20) + ADPG(19)
     *          - ARES(19)*AQ(20))
      Art(19) =  1.D0/(2.D0*Pi*Ar(19)*ALO(19))*AQ2
      Art(19) = 0.D0
C...
C...  Cerebral  [Arterial Segment 20]
C...
      AQ2 = AQ(20) - PQ(20)
      AQ2t = AQt(20) - PQt(20)
      AR2 = R2/ACAP(20)
      APt(20) = AQ2t*AR2 + AQ2/ACAP(20)
      Art(20) =  1.D0/(2.D0*Pi*Ar(20)*ALO(20))*AQ2
      Art(20) = 0.D0
C...
C...  Cerebral bed   [Peripheral Segment 20]
C...
      PQt(20) = 1.D0/AINERT(20)*(AP(20) - PP(20) + ADPG(20)
     *          - ARES(20)*PQ(20))
      VQ(20) = DMAX1((PP(20) - VP(20) - PQ(20)*PRA(20))/PRV(20),0.D0)
      PQ2 = PQ(20) - VQ(20)
      PPt(20) = PQt(20)*PRA(20) + PQ2/PCAP(20)
C...
C...  Venous Drainage from Brain  [Venous Segment 20]
C...
      VQt(20) = (PPt(20) - VPt(20) - PQt(20)*PRA(20))/PRV(20)
      VQ2 = VQ(20) - VQ(19)
      VQ2t = VQt(20) - VQt(19)
      VR2 = R2/VCAP(20)
      VPt(20) = VQ2t*VR2 + VQ2/VCAP(20)
      VQt(19) = 1.D0/VINERT(20)*(VP(20) - VP(19) + VDPG(20)
     *          - VRES(20)*VQ(19))
      Vrt(20) =  1.D0/(2.D0*Pi*Vr(20)*VLO(20))*VQ2
C...
C...  Midbrain   [Venous Segment 19]
C...
      PQOUT(18) = (PP(18) - VP(18) - PRA(18)*PQ(18))/PRV(18)
      PQOUTt = (PPt(18) - VPt(18) - PRA(18)*PQt(18))/PRV(18)
      VQ2 =  VQ(19) - (VQ(18) - PQOUT(18))
      VQ2t = VQt(19) - (VQt(18) - PQOUTt)
      VR2  = R2/VCAP(19)
      VPt(19) = VQ2t*VR2 + VQ2/VCAP(19)
      VQt(18) = 1.D0/VINERT(19)*(VP(19) - VP(18) + VDPG(19)
     *          - VRES(19)*(VQ(18) - PQOUT(18))) + PQOUTt
      Vrt(19) =  1.D0/(2.D0*Pi*Vr(19)*VLO(19))*VQ2
C...
C...  Opthalmic   [Venous Segment 18]
C...
      VQ2 =  VQ(18) - VQ(17)
      VQ2t = VQt(18) - VQt(17)
      VR2 = R2/VCAP(18)
      VPt(18) = VQ2t*VR2 + VQ2/VCAP(18)
      VQt(17) = 1.D0/VINERT(18)*(VP(18) - VP(17) + VDPG(18)
     *          - VRES(18)*VQ(17))
      Vrt(18) =  1.D0/(2.D0*Pi*Vr(18)*VLO(18))*VQ2
C...
C...  Upper Neck   [Venous Segment 17]
C...
```

```
      VQ2 =  VQ(17) - VQ(16)
      VQ2t = VQt(17) - VQt(16)
      VR2 = R2/VCAP(17)
      VPt(17) = VQ2t*VR2 + VQ2/VCAP(17)
      VQt(16) = 1.D0/VINERT(17)*(VP(17) - VP(16) + VDPG(17)
     *            - VRES(17)*VQ(16))
      Vrt(17) =  1.D0/(2.D0*Pi*Vr(17)*VL0(17))*VQ2
C...
C...   Lower Neck (Juglar)   [Venous Segment 16]
C...
      VQ2 =  VQ(16) - VQ(15)
      VQ2t = VQt(16) - VQt(15)
      VR2 = R2/VCAP(16)
      VPt(16) = VQ2t*VR2 + VQ2/VCAP(16)
      VQt(15) = 1.D0/VINERT(16)*(VP(16) - VP(15) + VDPG(16)
     *            - VRES(16)*VQ(15))
      Vrt(16) =  1.D0/(2.D0*Pi*Vr(16)*VL0(16))*VQ2
C...
C...   Subclavian   [Venous Segment 15]
C...
      VQ2 =  VQ(15) - VQ(2)
      VQ2t = VQt(15) - VQt(2)
      VR2 = R2/VCAP(15)
      VPt(15) = VQ2t*VR2 + VQ2/VCAP(15)
      VQt(2) = 1.D0/VINERT(15)*(VP(15) - VP(2) + VDPG(15)
     *            - VRES(15)*VQ(2))
      Vrt(15) =  1.D0/(2.D0*Pi*Vr(15)*VL0(15))*VQ2
C...
C...  Superior Vena Cava  [Venous Segment 2]
C...
        VQ2 =  VQ(2)  - VOQ(2)
        VQ2t = VQt(2) - VOQt(2)
        VR2 = R2/VCAP(2)
        VPt(2) = VQ2t*VR2 + VQ2/VCAP(2)
        Vrt(2) =  1.D0/(2.D0*Pi*Vr(2)*VL0(2))*VQ2
C...
C...  Form the derivatives for the peripheral beds
C...
C...   Opthalmic   [Peripheral Segment 18]
C...
      PQ2  = PQ(18) - PQOUT(18)
      PQt(18) = PPt(18)/PRA(18) - PQ2/(PRA(18)*PCAP(18))
C...
C...  Thorax and Coronaries [Peripheral Segment 3]
C...
      PQ2  = PQ(3)  - PQOUT(3)
      PQt(3) =  PPt(3)/PRA(3) - PQ2/(PRA(3)*PCAP(3))
C...
C...   Remainder of Peripheral Segments
C...
      i = 6
      DO WHILE(i .LE. IFOOTSEG - 1)
         IF ( POUT(i) .GT. 0 ) THEN
            PQ2  = PQ(i) - PQOUT(i)
            PQt(i) =  PPt(i)/PRA(i) - PQ2/(PRA(i)*PCAP(i))
         END IF
         i = i + 1
      END DO
      RETURN
      END
      SUBROUTINE PRINT(NI,NO6, NO7, NO8)
C...
C...  ODE COMMON
C...
C...   /Y/ time variables
```

Program CVSUBS.FOR

```
C...   /F/ time derivatives of variables
C...   /S/ spatial derivatives of variables
C...   /R/ & /I/ real and integer parameters required to define constants and
C...            define the spatial integration grid.
C...
      IMPLICIT DOUBLE PRECISION (A-H, O-Z)
      PARAMETER (NEQ = 20, NPSEG = 10)
      INTEGER NSTOP, NORUN, IP
      INTEGER*2 ALIN, ALOUT, VLIN, VLOUT, PVS, PIN, POUT
      DOUBLE PRECISION MM2PA, mu0, PPbed(NEQ)
      COMMON/T/         T,     NSTOP,     NORUN    ! Run Parameters
C...
C...  Arrays for segmental variables Pressure (P), Flow(Q), and radii (r)
C...
     1  /Y/    HP(4),      HQ(4),                  ! Heart's Chambers RA=1
     *       AP(NEQ),    AQ(NEQ),    Ar(NEQ),      ! Arterial P, Q, r
     *       VP(NEQ),    VQ(NEQ),    Vr(NEQ),      ! Venous P, Q, r
     *       VOP(3),     VOQ(3),                   ! Venous flows into heart
     *       PP(NEQ),    PQ(NEQ),                  ! Peripheral P, Qin, Qout
C...
C...  Time derivatives of the segmental variables: Pt, Qt, rt
C...
     2  /F/   HPt(4),     HQt(4),                     ! Heart's Chambers RA=1
     *      APt(NEQ),  AQt(NEQ),   Art(NEQ),          ! Arterial Pt, Qt, rt
     *      VPt(NEQ),  VQt(NEQ),   Vrt(NEQ),          ! Venous Pt, Qt, rt
     *       VOPt(3),   VOQt(3),                      ! Venous flows into heart
     *      PPt(NEQ),  PQt(NEQ),                      ! Peripheral Pt, QINt, QOUTt
C...
C... Parameters necessary to form the differential equations
C...
C...    Pi  -  3.14159...
C...    g0  -  9.80665  [m/sec^2]   earth's acceleration of gravity
C...  rho0  -  1050.    [kg/m^3]    density of whole blood (45% Hct)  | Assumed
C...   mu0  -  2.7      [cp]        viscosity of whole blood (45% Hct)| Constant
C...   D2R  -  pi/180   [radians/degree] scale factor
C...
     3  /R/  Pi, g0, rho0, mu0,  D2R, MM2PA, R2,        ! Constants
     *       ZAO(NEQ), ZAT(NEQ), ZVO(NEQ), ZVT(NEQ),    ! Arterial & Venous
     *       THETA(NEQ), HCAP(4), HVOL(4), HVR(4),                          ! Orientation angle
     *       ALO(NEQ), ArU(NEQ),  AE(NEQ), Ah(NEQ),     ! Arterial l, r, E, h
     *       ACAP(NEQ), ARES(NEQ), AINERT(NEQ),         ! Arterial Capacitance, resistance
     *       VLO(NEQ),  VrO(NEQ),  VrU(NEQ),            ! Venous l, r, rUNSTRESSED
     *       VCAPO(NEQ), VCAP(NEQ), VRES(NEQ), VINERT(NEQ), ! Venous Capacitance, resistance, inertance
     *       PRA(NEQ),  PRV(NEQ), PCAP(NEQ),            ! Peripheral Ra, Rv, C
     *       PINERT(NEQ), PVOL(NEQ),                    ! Peripheral I, V
     *       PAO(NEQ),  PVO(NEQ),                       ! Initial P conditions
     *       QAO(NEQ),  QVO(NEQ),                       ! Initial Q conditions
     *       AVOL(NEQ), VVOL(NEQ),                      ! A & V Volumes
     *       PEXT(NEQ),                                 ! Externally applied Pressure
     *       TO, GSTART, GMAX, TBRK1, TBRK2, TMAX, GFIN, ! G-Profile parameters
     *       Gz, ADPG(NEQ), VDPG(NEQ)                   ! Gz & Delta P from G
     4  /I/  IP, NDXPER(NPSEG),                         ! Peripheral Bed Indexes
     *       ALIN(NEQ), ALOUT(NEQ),                     ! Linkage Data arterial
     *       VLIN(NEQ), VLOUT(NEQ),                     ! Venous
     *       PIN(NEQ), POUT(NEQ),                     ! Peripheral
     *       PVS(NEQ),                                  ! Number parallel venous segments
     *       IFOOTSEG, IHEADSEG, IHEARTSEG              ! Foot, Head, & Heart seg nums
C...
C...
C...  PRINT A HEADING FOR THE NUMERICAL SOLUTION
      IP=IP+1
      IF(IP.EQ.1)WRITE(NO6,100)
      IF(IP.EQ.1)WRITE(NO7,100)
      IF(IP.EQ.1)WRITE(NO8,100)
      IF(IP.EQ.1)WRITE(*,1)
```

Program CVSUBS.FOR

```
C       WRITE(NO,2)
C       WRITE(NO,2) T
    1  FORMAT(2X,'T = ',F12.4,' sec'   )
C...
C...  PRINT THE SOLUTION
C...
      HQ(1) = VOQ(2) + VOQ(3)
      WRITE( *,22) T, Gz, THETA(1),(VP(k), k=1,NEQ), (VQ(k), k=1,NEQ)
      WRITE( *,22)  T, Gz, THETA(1),(HP(k), k=1,4), (HQ(k), k=1,4)
      WRITE(NO6,2) T, Gz, THETA(1),(AP(k), k=1,NEQ), (AQ(k), k=1,NEQ)
      WRITE(NO7,2) T, Gz, THETA(1),(VP(k), k=1,NEQ), (VQ(k), k=1,NEQ)
      i = 1
      DO WHILE (i .LE. NEQ)
        PPbed(i) = PP(i) - PQ(i)*PRA(i)
        i = i + 1
      END DO
      WRITE(NO8,2) T, Gz, THETA(1),(PPbed(k), k=1,NEQ),(PQ(k), k=1,NEQ)
    2 FORMAT(3F8.4,60(E12.4,1X))
   22 FORMAT(1X,3F8.4, /,12(5(E12.4,1X),1X,/))
  100 FORMAT(2X,' Time',3X,'     Gz     ','     Theta    ','Op Art Press',
     *' Op Art Flow',' Op Vn Press',' Op Vn Flow','Op PBed Pr  ')
      RETURN
      END
```

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BIBLIOGRAPHY

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