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TRANSIENT FISSION GAS BEHAVIOR IN URANIUM NITRIDE FUEL UNDER PROPOSED SPACE APPLICATIONS

A Dissertation

by

DANIEL LEE DEFOREST Captain, USAF 1991

Awarded PhD, Nuclear Engineering Texas A&M University

> Pages: 246 References: 126

In order to investigate whether fission gas swelling and release would be significant factors in a space based nuclear reactor operating under the Strategic Defense Initiative (SDI) program, the finite element program REDSTONE was developed to model 1-D, spherical geometry diffusion equations describing transient fission gas behavior in a single uranium nitride fuel grain. The equations characterized individual bubbles, limiting calculations to low temperatures, low burnups, or both. Instabilities in the bubble radii calculations forced additional constraints limiting the bubble sizes to minimum and maximum (equilibrium) radii.

Validity was checked against analytical solutions for internal consistency and against experimental studies for external agreement. These checks indicated REDSTONE was satisfactorily accounting for intragranular fission gas swelling, but other non-gaseous sources accounted for up to 1.5% additional swelling.

Calculations were performed to determine the swelling and gas release of a hypothetical SDI scenario, involving 10 years of low power operation at 10 kW followed by a 30 second rise to 1 MW and maintained for 15 minutes. The fuel pellet diametral increase remained less than 1% and release was negligible. Higher temperatures were used to see if operating parameter uncertainties would cause significantly different results. A significant shift to larger bubbles and larger swelling was indicated near 1800°K.

Poorly defined property values were varied and results compared. The largest effects were observed from the irradiation enhanced diffusion coefficient and the resolution parameter. The resolution parameter uncertainty had the largest potential effect on results.

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A Dissertation

by

DANIEL LEE DEFOREST

Submitted to the Office of Graduate Studies Texas A&M University in partial fulfillment of the requirement for the degree of

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December 1991

Major Subject: Nuclear Engineering

TRANSIENT FISSION GAS BEHAVIOR IN URANIUM NITRIDE FUEL UNDER PROPOSED SPACE APPLICATIONS

A Dissertation

by

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December 1991

ABSTRACT

Transient Fission Gas Behavior in Uranium Nitride Fuel Under Proposed Space Applications. (December 1991) Daniel Lee DeForest, B.S., Baylor University; M.S., Air Force Institute of Technology Chair of Advisory Committee: Dr. K.L. Peddicord

In order to investigate whether fission gas swelling and release would be significant factors in a space based nuclear reactor operating under the Strategic Defense Initiative (SDI) program, the finite element program REDSTONE (Routine for Evaluating Dynamic Swelling in Transient Operational Nuclear Environments) was developed to model the 1-D, spherical geometry diffusion equations describing transient fission gas behavior in a single uranium nitride fuel grain. The equations characterized individual bubbles, rather than bubble groupings. This limited calculations to those scenarios where low temperatures, low burnups, or both were present. Instabilities in the bubble radii calculations forced the implementation of additional constraints limiting the bubble sizes to minimum and maximum (equilibrium) radii.

The validity of REDSTONE calculations were checked against analytical solutions for internal consistency and against experimental studies for agreement with swelling and release results. These checks indicated that REDSTONE was satisfactorily accounting for the intragranular fission gas portion of swelling, but that other non-gaseous sources accounted for up to 1.5% additional swelling.

A calculation was performed to determine the swelling and gas release associated with a hypothetical SDI scenario. The SDI scenario involved 10 years of low power operation at 10 kW followed by a 30 second rise to 1 MW which was maintained for approximately 15 minutes. The results indicated that the fuel pellet diametral increase would remain less than 1% and release would be negligible, far below that needed for design failure. This same scenario was run at higher temperatures to see if uncertainties in the operating parameters could cause significantly different results. It was apparent that further investigations into the swelling behavior near 1800°K were needed, as results indicated a significant shift to larger bubbles at this temperature.

In an effort to understand the effect of property uncertainties on the solution, several of the properties were varied and the results compared. The largest effects were observed from changes in the irradiation enhanced diffusion coefficient and the resolution parameter. The resolution parameter uncertainty would have the largest potential effect on results, likely resulting in a reduction in the swelling and a corresponding increase in release.

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CHAPTER I INTRODUCTION

1.1 Justification

Proposed devices for use in the Strategic Defense Initiative (SDI) program will require at least a megawatt of power and will operate under stringent requirements. In addition to the harsh environmental requirements of space, the operational requirements will be quite severe. These latter requirements are the ability to survive in a low power 'housekeeping' mode for an extended period, perhaps up to ten years, with the ability to come to full power operation within seconds and maintain this power for a minimum of several minutes. Nuclear power is the most viable power option because of the constraints of power/weight ratio and hands-off operation.

Specific operational parameters for a multi-megawatt, SDI reactor are not publicly available. However, one can imagine a possible scenario where a reactor is sitting in 'housekeeping' mode at just enough power to keep the system functional, perhaps 10 kW electric. Upon command, a rapid increase to full power of 1 MW electric would then be executed, where the system would stay for several minutes. Conceivably, the ramp to full power might occur on the order of 30 seconds with the full power mode lasting for 15 minutes. The full power increase from 10 kW to 1MW in just 30 seconds could place severe thermal transients on the fuel. The high power operation could cause a large inventory of fission gas to coalesce into large bubbles inducing unwanted swelling of the fuel, or it could result in quick release of a large quantity of fission gas producing unwanted pressures in the fuel pin.

Style and format of this dissertation follow the Journal of Nuclear Materials.

One of the primary design constraints of an SDI nuclear system will be the amount of pin diameter increase allowed before the system is determined to have 'failed'. An increase in fuel pin diameter occurs either directly from expansion of the fuel against the cladding, or through excess internal pressure from fission gas which is released to the fuel-clad gap. The former case depends strongly on local conditions of the fuel pin since the fission gas is 'trapped' in the fuel at a specific location. In this case, the fission gas behavior both within the fuel grains (intragranular fission gas) and on the grain boundaries (intergranular fission gas) must be determined. The latter case depends less on local conditions since a large quantity of released fission gas will create open pathways out of the fuel and much of the gas that reaches the grain boundaries will be released to the gap where it mixes with gas released from other fuel pin regions. Here, intragranular fission gas behavior is of primary importance since it is implied by release that intergranular gas does not play a significant role. It is important to realize, however, that the actual behavior of the fuel will be some combination of the above two extremes. Nuclear fuel is usually designed to either release the fission gas to the gap or to accommodate the fission gas in the fuel. In either case, it is clear that the intragranular fission gas behavior must first be known in order to properly design the fuel. The importance of considering intergranular fission gas behavior is determined by the specific design of the fuel used.

The effects of thermally expanding fuel (fission gas swelling excluded) and the associated stresses and strains can be determined from macroscopic properties of the fuel and thus are easier to include in a design. However, the effects from fission gas, requiring knowledge of the microscopic properties of the fuel, are harder to determine. For the current scenario, the initial conditions of long term, low power operation followed by a severe increase in power by two orders of magnitude are unique to this system. A prediction of possible fission gas behavior in such a system has not been previously reported. Potentially, the fission gas could simply be released from the grain to grain boundaries which would become open avenues to the plenum either through cracks or interconnected grain boundary tunnels developed during the transient. On the other hand, the fission gas could be completely retained by the fuel, i.e. no cracks, and expansion of retained fission gas as bubbles could cause severe swelling of the fuel. The severity of this additional swelling would depend on where the fission gas resided during the transient. Intragranular fission gas would be present either as individual gas atoms or in very small bubbles. Intergranular fission gas tends to form much larger, lower density bubbles than found inside the grain. Grain boundaries are not as 'stiff' as the interior of the grain to expansion by excess bubble pressure. As a result, fission gas retained by the fuel on the grain boundaries will cause the most severe swelling of the fuel. When intergranular fission gas is retained, intragranular fission gas plays a lesser role in direct swelling of the fuel. However, knowledge of intragranular fission gas behavior is important as it is a feed for both intergranular bubbles and gas release. Thus, one must first analyze the intragranular fission gas behavior as a prerequisite to determining the effects from other fission gas processes.

1.2 Initial Indications

Scoping calculations based on the system and scenario as outlined above provided some insight into which processes would need to be included in modeling UN fuel. During the low power phase of the scenario, the fuel temperature appears to be cool enough that fission gas atoms are immobile in the lattice. Also, there is

only a small temperature gradient in the fuel due to the higher thermal conductivity of UN. Bubbles that do form will be from random diffusion with additional effects from the high density of fission gas atoms deposited over a ten year period. It is expected that during this phase the fission gas remains in small bubbles and single atoms and within the fuel grain. During the high power phase of the scenario, the fuel temperature and temperature gradient increases so that gas atoms and bubbles become more mobile. However, the time at full power is short enough that any bubble movement is severely limited. Initial calculations indicated that some release from the fuel grain and some bubble size increase might occur, but not to an extent that would cause fuel pin failure. In summary, initial indications were that the scenario being modeled would not cause large swelling or large release from the fuel grain, and that most of the fission gas would be retained in the grain. However, uncertainties in experimentally measured fuel properties for UN will cause corresponding uncertainties in any fission gas results calculated.

1.3 Use of Diffusion-Only Equations

As a result of scoping calculations, the computer program REDSTONE (<u>Routine for Evaluation of Dynamic Swelling in Transient Operational Nuclear</u> <u>Environments</u>), was written. This program is based on the diffusion equation for a 1-D spherical fuel grain. Effects from thermal gradients are included as additional coalescence terms to the diffusion equation and in some cases as simple release fractions to maintain spherical symmetry. This latter approach appears justified as the expected result is that the time at high power will be too short for significant gas movement to destroy spherical symmetry. This is fortunate since detailed modeling of asymmetric effects would cause the subsequent profiles within the fuel grain to

become non-symmetric and necessitate difficult and unwieldy 3-D modeling of the fuel grain.

1.4 Use of Finite Elements

Program REDSTONE was developed as a finite element code based on the 1-D non-linear diffusion equation in spherical coordinates. To be able to analyze the non-linear equation, the equations had to be incrementalized and solved through iteration. This technique has been used previously for structural mechanics. However, the type of non-linearity present in the diffusion equation used here is altogether different from that found in structural mechanics. In structural mechanics the non-linearities occur both as non-constant material properties and as cross products in the displacement gradients. In the formulation of the diffusion equation used here, the material properties are held constant (i.e. they are assumed not to be a function of fission gas atom concentration) and the non-linearities occur as cross products in the concentrations themselves, not in the concentration gradients. Thus the finite element formulation of the diffusion equation kich was used in REDSTONE is fundamentally different from that used in structural mechanics. It appears that this approach has not been reported previously.

1.5 Basic Structure of REDSTONE

The decision was made after some investigation into the formulation of the equations that the modeling of vacancies/interstitials within the fuel matrix and bubble radii should be decoupled from the calculation of bubble concentrations. It was found that inclusion of the vacancy/interstitial equations and bubble radii equations with concentration equations would cause such a high degree of non-linearity that an accurate finite element formulation using the methods applied

here could not be developed. The higher non-linearity would have reduced the concentration equations down to an essentially linear form destroying the benefits of using the method in the first place. As a result, the sequence of calculations in REDSTONE is to first calculate concentration changes, keeping vacancy/interstitial concentrations and bubble radii constant. This is followed by calculation of bulk vacancy and interstitial changes, keeping bubble concentrations and radii constant. After calculation of vacancy/interstitial concentrations, radii are recomputed and the entire process repeated until no additional iterations are required in the individual steps of bubble concentration calculation and radii calculation. Once convergence as described above has occurred, results are output, the time is incremented, and the process starts anew until the end of the problem. The calculation of bulk vacancy/interstitial concentrations and bubble radii are separated as the latter calculation is strictly linear and thus more efficient to calculate alone.

1.6 Objectives

The current work attempts to add insight to the modeling of fission gas within UN and answer some of the questions posed by the previously described scenario. The objectives of this study are:

- a) to investigate the modeling of fission gas behavior in uranium nitride (UN)
- b) to investigate the modeling of fission gas rate equations via a complete finite element formulation
- c) the determination of how much intragranular fission gas migrates to the grain boundaries and the amount of intragranular swelling
- d) the effect of uncertainties in the currently available property data

CHAPTER II SCOPING CALCULATIONS

2.1 Choice of System Parameters

2.1.1 Selection of a Nuclear System

Due to the lack of a current established design for an SDI nuclear system as well as the classified nature of the SDI program, a suitable substitute system needed to be chosen for this study. The most obvious substitute was the SP-100 nuclear reactor proposed for electric power requirements of space vehicles and satellites. This reactor system has an established design and it is conceivable that testing of SDI devices would utilize the SP-100 platform first [1]. Currently, the SP-100 is proposed as the most viable power option for systems requiring 10 kW to 1 MW of electric power. The figure-of-merit for failure of the fuel elements in this reactor has also been established. This limit is one percent diametral increase in the fuel element. Essential physical characteristics of SP-100 in the 1 MW configuration are listed in table 1.

2.1.2 Selection of a Nuclear Fuel

Because of the long lifetime required of the SP-100 reactor, as well as the need to have a compact, lightweight, and non-refuelable system, UN was chosen as the nuclear fuel for this reactor. The primary advantages of UN over UO_2 and UC include the higher density of UN, the much higher thermal conductivity of UN over UO_2 , and the expected lower fission gas swelling with burnup of UN. The higher density of UN allows for higher fission densities within the fuel and thus a smaller core (and mass) for producing the required amount of power. The higher thermal

Table 1

SP-100 Specification	S
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Number of Fuel Elements	3055
Fuel	UN
Enrichment	94%
% Theoretical Density	96%
Cladding	Nb1%Zr
Liner	Rhenium
Fuel Element Length	46.46 cm
Fuel Stack Length	7.62 cm
Cladding OD	0.607 cm
Cladding Thickness	0.055 cm
Liner Thickness	0.010 cm
 Fuel OD	0.456 cm

conductivity of UN results in lower temperature gradients in the fuel, thus allowing for higher operating temperatures (i.e. higher powers) and reducing internal thermal strain. The fission gas swelling directly affects the lifetime of the reactor and UN's lower propensity for swelling is principally derived from observations from the early 1970's. Researchers who have compared UO_2 with UN have attributed UN's swelling advantage to smaller thermal gradients [2]. Higher thermal gradients (and resulting higher fuel temperatures) in UO_2 introduce aggravations to swelling such as lenticular void growth, increased fission gas and bubble mobilities, and phase changes. Comparison of swelling data for UC with that of UN has suggested that, at identical temperatures, UC swells at a rate greater than both UN and UO_2 [2]. Fission gas swelling is potentially the largest of the factors which contribute to swelling of nuclear fuel, and is additionally the most complex and difficult to predict. Unfortunately, it is also one of the least characterized areas of UN.

2.2 Boundary Conditions

The first task in performing scoping calculations was to determine the temperature and pressure boundary conditions for the fuel element. The computer

code CENTAR (<u>Code for Extended Non-linear Transient Analysis of</u> Extraterrestrial <u>Reactor System</u>) [3] was used for this purpose. CENTAR was developed to model space reactor systems such as the SP-100. However, it does not contain a detailed fuel pin model. Instead a fuel pin is treated simply as a homogeneous material with a thermal conductivity and pin swelling determined from correlations, which are based on existing irradiation data. Because of this, their accuracy in unusual scenarios, such as the one being looked at here, is unknown. Fortunately, the boundary conditions as calculated in a systems code (i.e. coolant temperature and pressure) are not strong functions of the detailed structure of the fuel pins and thus should provide reasonably accurate input to a detailed fuel modeling code.

Calculations using CENTAR with the transient scenario considered here indicated that at 10 kW electric (200 kW thermal and the minimum needed to keep the system operational) the clad surface temperature was approximately 500°K, while at 1 MW (20 MW thermal) the surface temperature jumped to approximately 1500°K.

2.3 Fuel Temperature Profile

Using the calculated outer fuel pin temperatures, the steady-state heat conduction equation was solved analytically in order to find the centerline temperatures and thermal gradients in the fuel. The thermal conductivity was held constant to obtain the analytic solution.

The temperature increases from surface to centerline $(T_{CL} - T_S)$ were 5°K at T_s = 500°K and 300°K at T_s = 1500°K. The maximum thermal gradient is found at the fuel surface. These values are 38°K/cm at T_s =500°K and 2550°K/cm at

 $T_{\rm s}$ =1500°K. Based on these numbers one would expect that biased effects are not significant at 10 kW, but would need to be included at 1 MW. Since the low power operation will determine the pre-transient condition of the fuel, and because the transient time is so short, the following discussion will concentrate on effects present in the low power, low temperature phase of operation.

2.4 Fission Gas Source

In order to model a transient, the pre-transient condition must be known. The first step is the calculation of the amount of fission gas deposited in the grain at 200 kW thermal power over a ten year life. Dividing 200 kW by the total number of fuel pins in the core (3055), an average thermal rod power of 0.065 kW results. Assuming uniform flux within the fuel, the fission rate density will be 1.642×10^{12} fissions/sec-cm³. Using a grain diameter of 19 μ m, obtained from recent investigations involving SP-100 type fuel [4], one calculates that there are 185.7 fissions/sec within the grain. Over a ten year period this translates to 5.850×10^{10} fissions in a grain. Multiplying this by a stable fission gas yield of 0.25 fission gas atoms per fission [5], one finds that 1.462×10^{10} fission gas atoms are deposited in the grain over ten years, representing a density of 1.293×10^{20} fission gas atoms/cm³. If one assumes the gas atoms to be evenly dispersed as single gas atoms within the grain, the calculated distance between two gas atoms is 19.8×10^{-7} cm. Comparing this to the diameter of a xenon atom in the lattice (approximately 3×10^{-7} cm), one sees that local effects around each xenon atom could couple with other xenon atoms which could be highly disruptive to the lattice. If disruption is not expected then it must be assumed that fission gas is lost from the grain or that bubbles nucleate and grow from the deposited gas atoms. Since

disruption of the lattice is not expected, diffusion out of the grain and nucleation must be considered.

2.5 Steady-State Diffusion Out of the Grain

An indication of how much gas would be lost from the grain can be obtained by applying the steady-state diffusion equation to the single atoms deposited in the grain. Taking the boundary condition for diffusion calculations to be $C_{a} = 0$ at the grain boundary, one finds that the steady-state number of fission gas atoms contained within the grain is 3.22×10^{35} atoms. This number is much larger than the 1.462×10^{10} fission gas atoms deposited over ten years, so it is clear that a steady-state equilibrium is not attained. In fact, additional random walk calculations show that each xenon atom only moves a net distance of only $6.4 \times 10^{-9} \,\mu\text{m}$, far below the dimension of one unit cell in the lattice. Calculation of the homogeneous nucleation rate based on the fission gas concentration and thermal diffusivity indicates that this rate does not significantly alter the single gas atom concentration. Ignoring nucleation for the moment and applying the time-dependent form of the diffusion equation [6] one finds, as indicated above, that there is no significant release of fission gas to the grain boundary. Similarly, calculation of the thermal gradient effects show that biased movement of fission gas atoms is insignificant at this temperature. Scoping calculations thus indicate that once the xenon atoms are deposited, they are virtually 'frozen' in the lattice at low power. However, it should be noted that only thermal diffusion of fission gas was considered in the scoping calculations.

2.6 Heterogeneous Deposition and Nucleation

Although manufactured defects can act as heterogeneous nucleation sites, the primary source of defect sites will be the deposited atoms and fission damage created by fission. Accommodation of heterogeneous effects by some investigators has been done by including semi-empirical modifications to their models which account for fission damage and heterogeneous nucleation [7]. Other investigators have justified the use of a straight homogeneous approach. Unlike UO2, UN has a greater mobility of electrons which allow the energy transferred by a fission fragment to be deposited over a much larger region [8,9]. As a result, the energy transferred to a single lattice atom is lower and much less likely to impart enough energy to dislodge an atom. As a result, lattice damage from fission will be lower, resulting in fewer heterogeneous nucleation sites. A side effect of this property is that the increased electron mobility also prevents a significant thermal spike from being produced by the fission fragment. As a result, the resolution of bubbles should occur primarily via a Nelson knock-on method [10] versus a sputtering or trapping mechanism. Because of this difference in electrical properties, it would be questionable to simply transfer the UO_2 empirical modifications to UN. Unfortunately, modifications tailored to UN have not been developed due to the lack of appropriate experimental fission damage data. Because fission damage is lower in UN, heterogeneous nucleation will occur primarily through a finite probability that a deposited gas atom comes to rest next to an existing gas atom or bubble. Calculations of this probability involve the ratio of nearest neighbor sites for a specific bubble size to the total number of lattice sites available. Lattice damage from fission is not expected to provide a significant source of bubble nucleation.

2.7 Transient Regime

Similar calculations to those already discussed were performed for the transient regime. These calculations indicated that the transient time at high power will be too short for significant movement of fission gas. Thus, it is not expected that interactions between different sized bubbles will play a role during the higher power phase. However, the predicted amount of gas release is dependent on accurate modeling of intragranular fission gas behavior, which in turn depends on several properties of the material. Uncertainties in properties need to be considered in fuel design to determine acceptable safety margins.

CHAPTER III FISSION GAS MODELING IN UN

3.1. Historical Overview

Fission gas in nuclear fuels has been of interest to the nuclear community since the early 1960's. The theory of fission gas behavior has developed from a simple model of single gas atoms as solid fission products in pseudo-amorphous fuel [11, 12, 13] to the current state-of-the-art of detailed models describing fission gas behavior both within fuel grains and on grain edges and faces. The latter models attempted to account for the known factors which affect fission gas behavior, i.e. resolution from bubbles, heterogeneous and homogeneous bubble nucleation, bubble coalescence, bubble interlinkage, non-equilibrium bubbles, bubble mobility, nonstoichiometric effects, grain growth, impurities of solid fission products, bubble trapping at dislocations and defects, fuel cracking, fission enhancement of diffusion, etc. Unfortunately, the complex synergistic nature of fission gas behavior and the difficulty of studying the fuel at a microscopic level under operating conditions makes identifying and describing all the contributing and competing variables a monumental task.

3.1.1. Modeling Studies

Although some of the existing fuel modeling codes are based on empirical or semi-empirical correlations to describe fission gas swelling and release, these are only useful within the range of operational conditions used to define the correlations and lack the flexibility to perform exploratory analyses in new areas of interest. Exploratory analysis requires accurate mechanistic modeling codes based as much as possible on known theory. It is for this reason that program REDSTONE was

designed from a mechanistic approach. The theoretical basis of the mechanistic equations used in this work, and for the majority of current state-of-the-art mechanistic fuel modeling codes, dates back to Gruber [14] who, in 1966, first assembled a set of equations describing diffusion and interactive fission gas processes in nuclear fuel. Subsequent researchers extended and adapted Gruber's technique as their understanding about additional fission gas processes increased. Many of the extensions applied by subsequent investigators were situation specific and thus had limited usefulness for exploratory analysis. This resulting increase in complexity and specificity in Gruber's method appeared to continue until 1979, when emphasis seemed to shift from extension of Gruber's method to simplification of the method so it could be used efficiently in large integrated fuel modeling codes. The simplifications were primarily of two types, one involving collapsing the individual bubble groups into a single average bubble group and the other involving utilizing an effective diffusion coefficient for single gas atom calculations to account for all effects. These simplifications were developed and verified using calculations from the more detailed fuel modeling codes. For the most part, all were based on UO, fuel properties. Both Matthews and Wood [15,16] and Rest [17] have adapted Gruber's approach to their own codes, TRAFIC (TRAnsient Fuel Interpretative Code) and GRASS (Gas Release And Swelling Subroutine). Both these codes have evolved in similar directions and are representative of the mainstream of the current mechanistic approach to fission gas modeling. The processes considered in determining the effective diffusion coefficient or collapsing of bubble groups may or may not be the same in non-UO, fuel materials. The UO, simplifications might only be in agreement with the specific cases they were verified against, and thus would not be applicable to other fuel materials and/or operating histories. Before

simplifications for non-UO₂ fuel can be applied, initial modeling of fission gas behavior is best done by reversion to the more detailed equations of Gruber's method where the governing equations are more directly related to experimentally measured properties. This constraint affects the direction of the work here since the fuel of choice is UN, for which there are no detailed calculational studies.

3.1.2 Experimental Studies

As with the modeling developments, experimental studies involving nuclear fuels have concentrated essentially on UO_2 . In fact, both have, out of necessity, gone hand-in-hand. As a result, the modeling of UO_2 fuel behavior has advanced to a point that is quite detailed, but which cannot be duplicated for other fuel materials due to a lack of appropriate experimental data. Often, material properties necessary for applying current state-of-the-art models are estimated, usually from corresponding UO_2 data. For UN there is some support for substituting UO_2 properties as some researchers have suggested that certain aspects of fission gas behavior in UN are similar to those of UO_2 [2,18]. In addition to the lack of material property data, the lack of experimental studies involving non- UO_2 fuel materials. Both these problems have to be dealt with in the case of UN. The small amount of UN data available is further clouded by the fact that early investigators had problems in producing and maintaining pure, stoichiometric UN [19,20,21]. The state of the fuel being studied was not always well characterized.

3.1.3 Operational Studies

Studies which have dealt with operating scenarios involving power excursions, in addition to concentrating on UO₂, have generally considered fuel

which is operating initially at high powers (and thus high temperatures) and then experiences an overpower situation to very high temperatures where fuel restructuring is likely to occur. The controlling processes involved in these studies are different than those in the investigation here. Micrographs of UN fuel irradiated at relatively low temperatures have indicated that no restructuring of UN fuel occurs at low temperatures [19]. Predominantly low temperatures are expected and thus little or no fuel restructuring is expected to occur. This will result in a different pre-ramp condition of the fuel than in other studies. Matthews [15] indicates that the form of the intragranular gas at the start of a transient is decisive for subsequent release and swelling. No studies were found where the fuel was first operated for an extended period at the low temperatures used here and then experienced a severe overpower transient. Even if such studies were to exist, it is highly probable that they would have concentrated on the macroscopic fission gas release from a specific fuel design, probably using UO_2 , and not on the microscopic fission gas behavior of UN fuel.

3.2 Fission Gas Behavior Regimes

3.2.1 Intergranular Fission Gas

For intergranular (grain boundary) fission gas, modeling approaches have not congealed into any single mainstream method, possibly due to the greater effect initial manufacture and operating conditions have on open porosity and grain boundary dynamics. Early investigators considered the intragranular fission gas to be released once it reached the grain boundary, either immediately or once a specified percentage of the grain boundary had been filled with gas [22,23,24,25,26,27]. It was clear from accumulating experimental data, however,

that the dynamics of fission gas behavior on grain boundaries were much more complicated than these simple approaches suggested. In the 1970's, Tucker [28,29,30] looked at the energies involved in the grain boundary dynamics and was able to outline a detailed treatment of grain face and grain edge bubbles. Fission gas release occurred only after grain face bubbles had interconnected to the grain edges and grain edge bubbles were interconnected to the exterior or central void of the fuel. His approach has gained some acceptance and is the approach taken by Matthews [31]. Unlike the dominance of Gruber's method, however, Tucker's theories are not universally used for mechanistic modeling. At least one investigator has treated the grain boundary gas using diffusional rate equations similar to those used in Gruber's method [32]. Since intergranular fission gas was not modeled in the study here, no further elaboration of intergranular fission gas modeling will be made. If necessary, the amount of fission gas released will be determined using a percentage release criteria.

3.2.2 Intragranular Fission Gas

One of the first investigations into the theory of inert gas behavior in nuclear fuels was performed by Greenwood and Speight [33]. They based their theory on the assumptions that bubbles migrate by surface diffusion, there was no resolution of gas from bubbles, the gas in the bubble behaved according to the ideal gas law, and the bubbles were in equilibrium with the surrounding matrix. They also assumed the material to be a perfect, infinite crystal, and that the bubble distribution could be characterized by a mean bubble radius. Gruber [14] extended this work to eliminate the dependence on a mean bubble radius. He proposed using separate rate equations for single gas atoms and each bubble size. These rate equations included interaction terms between single gas atoms and bubbles to produce the bubble distribution. Other assumptions used by Greenwood and Speight remained assumptions for Gruber's work. Interaction terms included by Gruber (nucleation and coalescence) were based on colloidal coagulation work performed by Chandrasekhar [34]. The method of Gruber provided a basis that described the detailed bubble distribution, yet was easily adapted to specific cases by simply adding additional terms to the rate equations. The specific implementation of Gruber's method used here will be discussed in detail in the chapter covering governing processes.

3.3 Experimental Data on UN

3.3.1 Overall Status of UN Property Data

Available experimental data on UN, especially for the specific form of fuel proposed for the SP-100, is limited and generally not sufficient to fully characterize the microscopic processes occurring within the irradiated material. Advanced fuels, such as UN, were first considered as better alternatives to UO_2 , especially for LMFBR (Liquid Metal Fast Breeder Reactor) technology and possible applications in space [19,35,36,37], in the late 1950's to early 1960's. Early investigations of UN and UC, as well as the mixed fuels (U,Pu)N, and (U,Pu)C were performed in the mid to late 1960's. Unfortunately, work on these alternative fuels was overwhelmed by the data being collected on UO_2 and $(U,Pu)O_2$. As a consequence, investigations using UN dropped off in the early 1970's and only picked up again in the mid-1980's as the United States nuclear space program and SDI (Strategic Defense Initiative) came into the spotlight. Most of the experimental data on UN that were obtained in the 1970's utilized experimental techniques and equipment which were not as refined as today, and the effects of nonstoichiometry and fission damage in

UN were not well understood. Unfortunately, recent investigations utilizing UN have emphasized macroscopic observations (i.e. overall release and pin swelling) of particular fuel designs rather than the microscopic properties which would be useful for generalized fission gas modeling. It is difficult, if not impossible, to decipher the microscopic fission gas behavior from the experimental data in these more recent studies. Overall, experimental data available on UN are not sufficient for building comprehensive mechanistic models of microscopic fission gas behavior in UN fuel. As a result, some of the data necessary must be borrowed from UO_2 or UC.

Most of the experimental property data that was available on UN up to 1989 has been compiled into a database, as well as correlated with variables such as temperature and pressure, at Texas A&M University [38]. However, some of the primary data needed for modeling fission gas behavior is not present in this compilation and had to be gleaned from the few isolated investigations in the literature. Included in this data are the xenon diffusion coefficient, the surface diffusion coefficient, and the fission-enhanced diffusion coefficient. No measurements for some of the required data were found. Included were the surface energy in UN, the fission damage dependence of properties in UN, the effect of stoichiometry on UN properties, and the grain boundary energies of UN.

Where possible, the data used in REDSTONE was taken from the compiled database at Texas A&M. Other data were taken from the isolated investigations or taken from corresponding UC or UO_2 data.

3.3.2 Diffusion Coefficients

The diffusion coefficients of U, N, and the fission gases, as well as the surface diffusion coefficient of UN are absolutely essential to modeling fission gas

behavior. Each of the different types of diffusion coefficients are discussed in the following sub-sections.

3.3.2.1 Xe Diffusion Coefficient

Data for fission gas diffusion, whether in UN or other fuels, is typically taken to be the diffusion coefficient of xenon. This is a logical choice since most of the stable gaseous fission products consist of Xe (see table 2). Although some of the short-lived isotopes have half-lives on the order of the transient times considered here, the contribution of these to the total amount of gas already present in the grain at the beginning of the transient will be negligible. Unfortunately, both the Xe diffusion coefficient and the surface diffusion coefficient in UN are poorly documented.

Only a few measurements of the xenon diffusion coefficient in UN have been made. Melehan and Gates [20] measured the in-pile fission gas release from sintered uranium nitride in 1964. Irradiations and gas collection were performed at several temperatures ranging from 540°C to 1700°C. They noted that the fission gas release rate up to 800°C was independent of temperature. This was attributed to a recoil knockout - evaporation release mechanism. Above 800°C, the release rate followed an Arrhenius behavior described by the equation

$$D_{\rm Xe} = 4 \times 10^{-6} \exp\left(-\frac{4.5 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \tag{1}$$

where the activation energy is 4.5×10^{-12} ergs. Their analysis of the release data was based on the theoretical work of Beck [39]. Although an equivalent sphere approach was used, the size of the spheres was determined strictly from the physical dimensions of the fuel element, not from any analysis of grain size. Also, no

Isotope	Half-Life	Percent Yield	
¹³¹ Xe	Stable	3.2	
¹³² Xe	Stable	4.7	
¹³³ Xe	5.3 day	6.6	
¹³⁴ Xe	Stable	6.6	
¹³⁵ Xe	9.2 hr	5.5	
¹³⁶ Xe	Stable	5.9	
	Total stable xenon yields		
⁸³ Kr	Stable	0.4	
⁸⁴ Kr	Stable	0.85	
⁸⁵ Kr	Stable(10.6 yrs)	0.15	
^{85m} Kr	4.4 hr	1.3	
⁸⁶ Kr	Stable	1.4	
⁸⁷ Kr	78 min	2.5	
	0.0.1	3.5	
⁸⁸ Kr	2.8 hr	3.5	

Table 2

Isotopes of Krypton and Xenon Released in Fission [5]

correction was performed to account for interactions of gas atoms with bubbles or the effect from nonstoichiometric fuel. Post-irradiation examination of the fuel did not reveal evidence of density changes as would be expected from decomposition of UN. However, since a nitrogen overpressure was not maintained, it would be expected that some degradation of the fuel occurred and that the post-irradiation examination was not able to detect it. Any decomposition would affect the resulting calculated diffusion coefficient.

About the same time, Biddle [40] measured the post-irradiation emission of xenon from uranium carbonitride powders containing varying amounts of carbon. The equivalent sphere model developed by Booth [41] was utilized, with the equivalent sphere size being determined from krypton adsorption measurements onto the surface areas. Biddle's analysis did not account for gas atom interactions and no effort appeared to be made to keep the UN stoichiometric during measurement. Resulting diffusion equations for two different surface areas were

$$D_{\rm Xe} = 3 \times 10^{-8} \exp\left(-\frac{4.0 \times 10^{-12}}{kT}\right)$$
 cm²/s: Surface area = 1850 cm²/g (2)
and

$$D_{\rm Xe} = 8 \times 10^{-8} \exp\left(-\frac{5.6 \times 10^{-12}}{kT}\right)$$
 cm²/s: Surface area = 1000 cm²/g.(3)

These two results appear markedly different. However, results presented for all the fuel materials seem questionable since values for the carbonitrides exhibit just as wide a range with no observable trend as a function of carbon content.

Measurements of xenon diffusion in single crystals of UN were performed by Oi [42]. The evaluation of the diffusion coefficient was based on the measured release fraction and the surface/volume ratio of the bulk crystals. No allowance was made for gas atom interaction and no attempt was made to maintain stoichiometric UN during measurements. The resulting diffusion coefficient was

$$D_{\rm Xe} = 2.05 \times 10^{-4} \exp\left(-\frac{3.57 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \,.$$
 (4)

This equation produces values higher than those found by other investigators. This may be the result of a lack of traps for gas atoms in a single crystal. Oi also mentions that the scatter in their data was considerable.

Work carried out in 1970 at BMI [43] measured fission gas release from 97 percent dense compacted UN which had been neutron activated to produce fissions. The resulting data were analyzed using a Hurst [44] model (see table 3). The Hurst model approximates the effects from fission gas precipitation. A nitrogen overpressure was maintained during analysis to keep the UN stoichiometric. The diffusion coefficient for UN was found to be

	Estimated Diffusion Coefficient, cm ² /sec				
Fue	1400°C	1600°C			
UC	3×10 ⁻¹²	8×10 ⁻¹¹			
UN	2×10^{-13}	2×10 ⁻¹²			
UO	4×10 ⁻¹³	2×10 ⁻¹¹			

 Table 3

 Estimated ¹³³Xe Diffusion Coefficients for UC, UN, and UO,

$$D_{\rm Xe} = 4.65 \times 10^{-4} \exp\left(-\frac{4.9796 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \,.$$
 (5)

Ritzman [19] notes that diffusion coefficients calculated using the Hurst model are several orders of magnitude higher than values previously reported due to the inclusion of precipitation of gas atoms. This may explain the lower values obtained by Biddle [40].

Later diffusion work at NASA [45] used a sweep gas facility to measure in-pile krypton release rates from small UN specimens irradiated to eight percent burnup and at high fission rate density. They correlated their data with predictions of a fission gas release model which included atomic diffusion, fission-enhanced diffusion, and direct recoil. Calculation of any one of these contributors depended on the accuracy of the other two. Because of the form of their equation, they were required to calculate the effective surface area for release. This was accomplished by assuming complete interconnection of grain boundary porosity. No correction was made for gas atom interaction, bubble movement, or biased motion, even though bubbles were clearly evident in post-irradiation ana¹ sis. As a result, the calculated diffusion equation represents overall release from the fuel irregardless of species. The diffusional contributors to the overall gas release were modeled with the steady-state solution to the diffusion equation in rectangular coordinates, not with an equivalent sphere model as used previously. The resulting diffusion equation they obtained for atomic diffusion was

$$D_{\rm Kr} = 5.6 \times 10^{-8} \exp\left(-\frac{1.08 \times 10^{-11}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \tag{6}$$

The high activation energy present above may be a direct result of neglecting contributions from bubbles and gas atom traps such as grain boundaries (complete interconnection was assumed). Also, no attempt was made to keep the UN stoichiometric, and the small size of the specimens may have caused normally insignificant processes to interfere with release. In addition, it should be noted that the diffusion equation presented above describes krypton diffusion, not that of xenon. Thus some difference might be expected.

Further work at NASA [46] extended the above analysis to include contributions from bubble diffusion. However, other criticisms of the initial study remained. Their adjusted atomic diffusion equation for krypton was

$$D_{\rm Kr} = 2.4 \times 10^{-10} \exp\left(-\frac{2.60 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \tag{7}$$

It is unclear if the inclusion of bubble motion was enough to effect the large changes observed in the diffusion equation. The latter activation energy is certainly consistent with those of other investigators. However, the pre-exponential factor appears to be far too low. Differences between the two studies were not discussed sufficiently to explain the change in diffusion equations.

The dependence of diffusion coefficients on the amount of fission damage present has been used as one explanation of the differences obtained by various investigators. Experiments with UC have indicated a definite dependence of the xenon diffusion coefficient on fission damage [47]. Unfortunately, a similar trend cannot be applied to UN as the amount of fission damage in the preceding study was not specified and fission damage work with UN has not yet been performed. Fission damage dependence is not included in program REDSTONE.

A review of the various fission gas diffusion equations indicates that the latter results at BMI appear to be the most reliable. Not only were steps taken to preserve the integrity of the UN, but account was made for fission gas traps not included by other investigators. Additionally, the form of the fuel used at BMI (sintered, high density, stoichiometric with some fission damage from neutron activation) was similar to that proposed for space reactors. The diffusion equation used in REDSTONE is that derived from the data at BMI.

3.3.2.2 Irradiation Enhanced Diffusion

Irradiation enhanced diffusion of fission gas in nuclear fuels is not well characterized, even for UO₂. Although Dienes and Damask [48] have looked at this problem theoretically, in practice irradiation enhanced diffusion coefficients are obtained strictly from experiment. In these cases, irradiation enhanced diffusion is used to explain an athermal release of fission gas at lower temperatures. For UO₂, the critical temperature for irradiation enhanced diffusion is around 1000°C [49,50]. For UN, the critical temperature appears to be lower around 800°C [20]. Cornell [51] derived an irradiation enhanced diffusion based on experimental observations of bubble sizes and concentrations, and obtained an expression that depended on the fission rate and resolution rate. The irradiation enhanced diffusion coefficient is usually expressed in terms of a constant and the fission rate, F. Typical values for UO, range from [49,52]

 $D_{\rm E} = 2.0 \times 10^{-29} \dot{F}$ to $2.0 \times 10^{-31} \dot{F}$ cm² / s

26

(8)

where F is the fission rate. For UN, investigators at NASA (National Aeronautical and Space Administration) [46] appear to be the only ones to have calculated an irradiation enhanced diffusion rate. The value they found for UN was

$$D_{\rm E} = 8.22 \times 10^{-31} \dot{F} \,\,{\rm cm}^2 \,/\,{\rm s}$$
 (9)

This study was also discussed in the section on fission gas diffusion which involved measured krypton releases using sweep gas over small specimens of UN. The results were fit to an equation which included contributions from irradiation enhanced diffusion, atomic diffusion, bubble diffusion, and direct recoil. Although it was mentioned that these contributions were interdependent, evaluation of the irradiation enhanced diffusion coefficient was accomplished at low temperatures where other diffusional processes were negligible. The decrease in calculated irradiation enhanced diffusion from the UO_2 value can be explained from the lower amount of fission damage expected in UN [49,53]. Since no other source for irradiation enhanced diffusion in UN exists, the value obtained by NASA is used in REDSTONE.

3.3.2.3 Bubble Diffusion Coefficients

Three mechanisms have been proposed to explain bubble mobilities: vapor transport, volume diffusion, and surface diffusion. Vapor transport occurs from the evaporation of atoms on the hot side of a bubble and deposition on the cold side. Since surface atoms must break completely away from the surface, high activation energies are present, and thus high temperatures are required. For the low temperatures considered here, vapor transport should be negligible. Experimental evidence for UO_2 , where the driving forces and temperatures from temperature

gradients are much higher, indicates that this assumption is valid up to moderately high temperatures [54,55].

Volume diffusion occurs from the bulk movement of atoms in the matrix around the bubble. Calculations on the relative effects between surface diffusion and volume diffusion have shown that only when bubbles are larger than about 1mm in diameter [15] is volume diffusion significant. Since only small bubbles are expected, volume diffusion is not considered to be a factor in the study here, which leaves only surface diffusion to explain bubble mobility. By far the most widely accepted model for bubble mobility is the surface diffusion mechanism. Utilizing the surface diffusion model, the bubble diffusion coefficient is

$$D_{\rm n} = \frac{3\lambda\Omega}{2\pi R_{\rm n}^4} D_{\rm s} \tag{10}$$

where D_n = bubble diffusion coefficient, D_s = surface diffusion coefficient, λ = atomic spacing, Ω = atomic volume, and R_n = bubble radius.

The surface diffusion coefficient for UN was measured by Weaver in 1968 [56] using helium implanted UN. After annealing, bubble sizes and migration distances up a thermal gradient were measured through replication electron microscopy at temperatures ranging from 1258°K to 1858°K and thermal gradients from 75°K/cm to 880°K/cm. Although their original data were not presented, they did calculate a temperature dependent surface diffusion coefficient of

$$D_{\rm s} = 1.92 \times 10^3 \exp\left(-\frac{2.93 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s}$$
 (11)

Also observed was that coalescence of bubbles did not occur as the bubbles traveled up the thermal gradients. Weaver attributed this to the existence of large stress fields surrounding nonequilibrium bubbles. This latter phenomena has been

theoretically investigated by Ronchi [57]. The surface diffusion equation obtained by Weaver is used in program REDSTONE.

If one uses the above equation to calculate a bubble diffusion coefficient for a bubble the size of one xenon atom, one gets a value several orders of magnitude higher than the xenon diffusion coefficient. Even further, Cornell [51] through observations of irradiated fuel postulated that very small bubbles are immobile due to being trapped at dislocations, and thus cannot be described by a diffusion coefficient as given above. However, Nichols [58] looked at much of the data on bubble diffusion and concluded that small bubbles move at rates far greater than those assumed by Cornell, though still not at rates predicted from theory. Lower small bubble mobility is probably a factor since small bubbles in UO, have been observed to be faceted rather than spherical [59, 60], and thus would not be governed by the same enthalpies as those of larger, purely surface-diffusion controlled bubbles. This presents a complication, however, since it is general practice to treat all bubbles as being controlled by the same mechanism. So that small bubbles are accurately modeled, bubble diffusion coefficients should exhibit a smooth transition from single xenon atoms to the larger, surface diffusion controlled bubbles. Various theories of bubble mobility, some based on other diffusion mechanisms, have been proposed to explain experimental observations and all claim good agreement with the data [61, 62]. No predominant theory has emerged. Gruber [63] and Chien [32] have adjusted small bubble diffusion coefficients by considering the surface atom jump distance when it is greater than the circumference of the bubble. Ronchi [64] adjusts bubble diffusion coefficients based on a transition from xenon atom to a bubble size large enough so that atomic strains around the bubble are completely relaxed. Because both methods give similar

results, Gruber's approach is applied in REDSTONE due to its easier implementation.

Gruber's approach was to apply a 'rolling stone' model for surface diffusion. Robertson [65] has postulated that this model is a more accurate description of surface diffusion. The classic model of surface diffusion is that it occurs one interatomic distance at a time via the movement of vacancies on the surface. In the 'rolling stone' model, the movement of surface atoms occurs when an atom pops out of the surface and 'rolls' along the surface to an existing vacancy. The distance of the 'roll', i.e. the mean jump distance, will span 10-1000 interatomic distances. Gruber shows that this model produces results consistent with existing data.

To illustrate the approach, a bubble of radius ' R_n ' is assumed with a surface atom mean jump distance of λ_s , assumed to be constant and independent of surface orientation and curvature. The net movement of the bubble, λ_n , can be calculated from simple geometry from the arc length, λ_s , and the bubble radius, R_n , i.e.

$$\lambda_{\rm n} = 2R_{\rm n}\sin\left(\frac{\lambda_{\rm s}}{2R_{\rm n}}\right) \tag{12}$$

For $\lambda_s << \pi R_n$, the above becomes $\lambda_n = \lambda_s$. For $\lambda_s > \pi R_n$, λ_n is periodic. Because this result yields unrealistic behavior, Gruber assumed for $R_n < \lambda_s / \pi$ that $\lambda_n = 2R_n$. The correction for bubble diffusion can then be represented by

$$D_{\rm n}^{\rm new} = D_{\rm n} \left(\frac{\lambda_{\rm n}}{\lambda_{\rm s}}\right)^2 \tag{13}$$

Substituting the appropriate expressions for the correction factor into the above equation gives

$$D_{\rm n}^{\rm new} = \frac{6\lambda\Omega D_{\rm s}}{\pi\lambda_{\rm s}^2 R_{\rm n}^2} \sin^2\left(\frac{\lambda_{\rm s}}{2R_{\rm n}}\right) \qquad \qquad R_{\rm n} > \frac{\lambda_{\rm s}}{\pi} \qquad (14)$$

The only additional unknown is the jump distance, λ_s . Gruber calculated λ_s by applying his modifications to the data of Gulden [61], which indicated that bubbles smaller than 37 Å appeared to follow an R_n^2 dependence rather than the R_n^4 dependence expected from surface diffusion. Gruber was able to predict the trends in Gulden's data using surface diffusion and his modifications. He was also able to determine λ_s from the data to be approximately 112 Å. Since data on UN to perform a similar analysis are not available, it is assumed in REDSTONE that $\lambda_s = 112$ Å for UN.

3.3.2.4 Self-Diffusion Coefficients

Experimental data on U and N diffusion in UN are more abundant than for fission gas diffusion. The available data have been surveyed and correlated with nitrogen partial pressure, temperature, and grain size in a database recently compiled at Texas A&M University [38]. Since the work involved in compiling the database considered data quality and experiment reliability, it was decided to simply use the correlated values from the Texas A&M University database. An extensive review of the available self-diffusion data will not be presented here. Interested readers are referred to the above reference. The correlated equations for self-diffusion of U and N in UN are

$$D_{\rm U} = 2.2146 \times 10^{-11} P^{0.6414} \exp\left(-\frac{1.1029 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \tag{16}$$

and

$$D_{\rm N} = 2.251 \times 10^{-5} P^{0.4134} g^{-0.737 + 2.179 \times 10^{-4} T} \exp\left(-\frac{2.6527 \times 10^{-12}}{kT}\right) \quad {\rm cm}^2 \,/\,{\rm s} \quad (17)$$

where P = nitrogen partial pressure (atm), T = temperature (K), and g = average grain size (μ m).

3.3.2.5 Vacancy/Interstitial Diffusion Coefficients

The diffusion of vacancies and interstitials is poorly documented for nuclear fuels. Previous investigators adopted values from other materials. Because it is the vacancies that cause bubble growth, several investigators who have considered the effects of vacancies have not included interstitials. Further, only a few investigators who have incorporated vacancy effects have published the actual diffusion data used. Additionally, vacancy effects are often incorporated only implicitly by utilizing creep properties of the fuel, not by a set of coupled rate equations. As a result, there is slightly more documentation concerning vacancies than interstitials.

For interstitials, both Hayns [66] and Dienes and Damask [48] use the same value for the migration energy, $Q_i^m = 3.20 \times 10^{-13}$ ergs, where Q_i^m is the energy of migration for interstitials. Hayns took his value from austenitic steel while Dienes and Damask took theirs from a representative metal 'such as copper'. Since the two values agree, it may be that Dienes and Damask's representative metal was not all copper, but rather a conglomeration of data available. This is a reasonable hypothesis since Dienes and Damask's work predates the others discussed by

14years. Hayns used a pre-exponential factor of 10⁻³, while Dienes and Damask used a theoretical approach to arrive at 10^{-2} . However, several values needed to arrive at the latter value had to be contrived. Thus, it cannot be automatically assumed that the theoretical value is better than other values for which the origin is unknown. Straalsund and Guthrie [67] assumed a pre-exponential factor of one for all their diffusion equations and used an interstitial migration energy value, reportedly from austenitic steel, which was different from the above, i.e. $Q_i^{m}=2.07 \times 10^{-13}$ ergs. The reason for the discrepancy in values is unknown. However, they mention that many of the parameters needed are highly uncertain, and thus some of the data involves reasonable estimates. Because it cannot be determined which values best apply to UN, and given the agreement between Hayns, and Dienes and Damask, the diffusion of interstitials in REDSTONE was chosen to be represented by

$$D_{\rm i} = 10^{-3} \exp\left(-\frac{3.20 \times 10^{-13}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \,.$$
 (18)

None of the above works which considered interstitials were concerned with fission gas modeling. Esteves, et al. [68] explicitly considered vacancies (but not interstitials) in modeling fission gas behavior in UO_2 . Their form of the diffusion equation for vacancies was derived by taking the pre-exponential factor from uranium self-diffusion and a vacancy migration energy equal to the vacancy formation energy, i.e.

$$D_{\rm v} = 4 \times 10^{-7} \exp\left(-\frac{2.44 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s}$$
 (19)

where $Q_v^m = Q_v^f = 2.44 \times 10^{-12}$ ergs. The energy of formation was taken as one-half the activation energy for self-diffusion. The basis for their pre-exponential factor is questionable since Olander [5] notes that

$$D_{\rm v} = D_{\rm U} C_{\rm v} \Omega \ . \tag{20}$$

Thus, one would not expect the self-diffusion pre-exponential factor and the vacancy pre-exponential factor to be the same. Additionally, it would be expected, as is the case with the other investigators reviewed, that the migration energy would be approximately 80% to 90% of the formation energy. Other investigators took their values from other metals. Dienes and Damask [48] again used values from a 'representative metal'. Their theoretically based pre-exponential factor was one and the vacancy migration was $Q_v^m = 1.28 \times 10^{-12}$ ergs. As will be seen, this value is much lower than others. Because of the previously mentioned uncertainties in the source of Dienes and Damask's data, their values for vacancy diffusion were not considered for use in REDSTONE. Straalsund and Guthrie [67], Hayns [66], and Brailsford and Bullough [69] all use values derived from steel. Straalsund and Guthrie assumed a pre-exponential factor of one, and used a migration energy of $Q_v^{m}=2.05 \times 10^{-12}$ ergs. Hayns, and Brailsford and Bullough both used a pre-exponential factor of 0.6. However their migration energies were slightly different with Hayns using $Q_v^m = 2.24 \times 10^{-12}$ ergs, and Brailsford and Bullough using $Q_v^{\rm m} = 2.08 \times 10^{-12}$ ergs. The discrepancy between these last three migration energies could not be resolved, although it is possible that the value used by Straalsund and Guthrie involves a 'reasonable estimate'.

The vacancy migration energy used in REDSTONE is an average of these three energies. The pre-exponential factor was taken as 0.6, i.e.

$$D_{\rm v} = 0.6 \exp\left(-\frac{2.12 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^2 \,/\,{\rm s} \,.$$
 (21)

3.3.3 Miscellaneous Data

Non-diffusional experimental data is also required to adequately model fission gas behavior. The more important ones are discussed below.

3.3.3.1 Thermal Equilibrium Concentrations

The equilibrium concentrations of vacancies and interstitials, although insignificant in an irradiation environment, are needed to specify the boundary conditions of bubbles, dislocations, and grain boundaries. The equilibrium concentrations are governed through thermal emission via

$$C_{\rm v}^{\rm e} = \frac{1}{\Omega} \exp\left(-\frac{Q_{\rm v}^{\rm f}}{kT}\right) \tag{22}$$

and

$$C_{i}^{e} = \frac{1}{\Omega} \exp\left(-\frac{Q_{i}^{f}}{kT}\right)$$
(23)

where Ω = atomic volume, Q_v^f = vacancy energy of formation, and Q_i^f = interstitial energy of formation. As with diffusion data, the energies of formation are poorly documented. This is particularly true for interstitials. Olander [5] notes that, unlike vacancy formation, a variety of interstitial sites are available and thus it is harder to determine the interstitial energy of formation. Again, investigators have used data available from similar metals. Straalsund and Guthrie [67] used a value from austenitic steel, $Q_i^f = 6.9 \times 10^{-12}$ ergs, while Dienes and Damask [48] used a value from their 'representative metal', $Q_i^f = 6.4 \times 10^{-12}$ ergs. Olander notes further that the energy of formation for interstitials is probably higher than that for vacancies in all solids. In practice for nuclear fuels, the equilibrium concentration of interstitials is taken as zero due to the high formation energy [5,48,70,71]. This same practice is used in REDSTONE, i.e. $C_i^e = 0$.

For vacancies, the equilibrium concentration is retained. Hayns [66] and Brailsford and Bullough [69] used the same values from steel, $Q_v^f = 2.56 \times 10^{-12}$ ergs. Straalsund and Guthrie [67] used energies of formation in the range $Q_v^f = 2.6 \times 10^{-12}$ to 2.8×10^{-12} ergs. Dienes and Damask [48] used values from their 'representative metal', $Q_v^f = 1.92 \times 10^{-12}$ ergs. Like their value of the vacancy migration energy, this value is much lower than that used by other investigators. Because of this discrepancy, this value was not considered for use in REDSTONE. As discussed earlier, Esteves, et al. [68] took half of the activation energy for self-diffusion in UO_2 as the formation energy for vacancies [72], i.e. $Q_v^f = 2.44 \times 10^{-12}$ ergs. Another fission gas modeling work, Ting, et al. [73], modeled UC using a vacancy formation energy provided by DonnerandSchiile[74], $Q_v^f = 2.71 \times 10^{-12}$ ergs.

All the values above, except for that of Dienes and Damask, fall in the same general range. Since one value does not appear any more applicable than another, the vacancy energy of formation used in REDSTONE is an average of the above values, with the exception of Dienes and Damask's value, i.e. $Q_v^f = 2.60 \times 10^{-12}$ ergs, with

$$C_{\rm v}^{\rm e} = \frac{1}{\Omega} \exp\left(-\frac{2.60 \times 10^{-12}}{kT}\right) \qquad {\rm cm}^{-3} .$$
 (24)

3.3.3.2 Surface Energy

The surface energy is important for determining bubble sizes and excess pressures. No available data on the surface energy of UN were found. Surface energy data is available concerning the other two nuclear fuels, UO, and UC. An early experimental study [75] found the free surface energy of UO_2 to be 640 ergs/cm² and the free surface energy of UC to be 1000 ergs/cm². These numbers are very approximate, with stated uncertainties on the order of 20 to 30 percent. The free surface energy of UO₂ was calculated from theory by Benson [76] to be 1030 erg/cm². Warner and Nichols [77] estimated the surface energy to be approximately 1000 ergs/cm². About the same time, Eberhart [78] found critical surface energies of UO₂ to lie between 600 and 1650 ergs/cm² depending on whether an argon or hydrogen atmosphere was present. Later, Maiya [79] performed experiments of the growth of grain boundary grooves and determined an average free surface energy of 626 ergs/cm^2 . Clearly, a consistent agreement on the value of free surface energy in UO₂ has not been reached. Most often, the value used in fuel modeling codes is 1000 ergs/cm². For UC, reported surface energy values have varied just as much. Initial studies set the free surface energy at 730ergs/cm² [80] to 780 ergs/cm² [81]. However, further work indicated that the free surface energy of UC was as high as 2000 erg/cm² [82,83]. Although some fuel modeling codes have used the lower values for UC, the spread of available data suggests that the use of 1000 ergs/cm² would not be unreasonable. Since no reported data for free surface energy of UN have been found, and consistent values are not available for either UO, or UC, the value for free surface energy of UN used in REDSTONE is 1000 ergs/cm².

3.3.3.3 Lattice Parameter

As with self-diffusion, the lattice parameter of UN has also been correlated with temperature at Texas A&M University [38]. The equation $a = 4.879 + 3.264 \times 10^{-5}T + 6.889 \times 10^{-9}T^2$ Å (25) is used in REDSTONE where T = temperature (K).

3.3.3.4 Yield Stress

The yield stress is used in REDSTONE to determine the minimum sizes of bubbles. The yield stress for UN was approximated from the stress-strain curves of Werner and Blank [84]. Values consistent with the plastic behavior were read from the curves and then correlated with a least squares fit. The resulting correlation is $\sigma_y = -9.256 \times 10^8 T + 1.735 \times 10^{12}$ Pa. (26)

The values produced with such a correlation are very crude. Fortunately, bubble sizes change very little with large changes in pressure near this stress. Thus, inaccuracies in the yield stress should have minimal effect on the program results. The main effect of the yield stress is to keep the bubble sizes from becoming too small.

3.3.3.5 Thermal Conductivity

The thermal conductivity of UN has been correlated with temperature and porosity at Texas A&M University [38]. This correlation was used in determining the temperature profile in the fuel during scoping calculations. The thermal conductivity of porous UN is given by

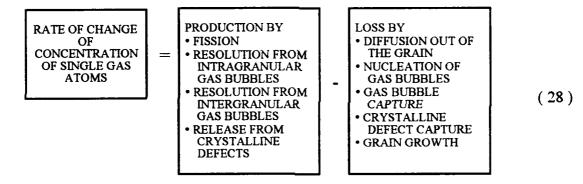
 $k_{\rm p} = 1.864T^{0.0361} \exp(-2.14P)$ W/m-k (27)

where T = temperature (K) and P = volume fraction porosity.

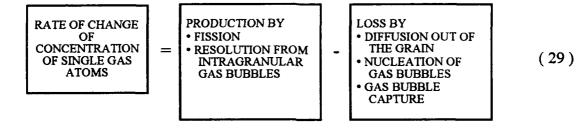
CHAPTER IV GOVERNING EQUATIONS AND TERMS

4.1 Governing Processes

The equation which is the basis of program REDSTONE, the diffusion equation, can be considered a balance of production and loss rates of individual species, be they gas atoms, gas bubbles, vacancies, or interstitials. Considering first individual gas atoms, a balance of production and losses within the grain typically would consider the following:



For UN fuel and the operating scenario to be considered here, some of the above processes can be eliminated from consideration. Based on scoping calculations, the amount of intergranular gas would be so small that resolution effects from these bubbles would be negligible. Assuming no grain boundary resolution also provides for the worst case situation where diffusion out of the grain is maximized. Within the UN fuel grain, the effect of crystalline defects on gas atoms will be small, in the case here assumed negligible, due to the lower amount of fission damage expected in UN [9]. The dominance of fission gas bubbles will provide for the major capture defect [85]. The effect of fission damage defects is somewhat accounted for as the Xe diffusion coefficient used was measured in irradiated, fission damaged material [43]. Loss by grain growth can also be eliminated, since the temperatures of interest during the long term operation are too low for significant grain growth to occur [20,37]. During the short term at high power, the time is too short for any significant grain growth. After eliminating the above terms, a simplified equation can be written:



Biased effects are implicitly included in the above. Modeling of biased migration effects can likewise be simplified since biased migration would be minimal in UN at low temperatures and at high temperatures the time available severely limits atom and bubble movement. Specific simplifications to biased effects will be discussed in sections dealing with those effects.

For gas bubbles, a similar production-loss equation can be written:

RATE OF CHANGE OF CONCENTRATION OF GAS BUBBLES OF 'N' ATOMS	PRODUCTION BY • FISSION • NUCLEATION OR COALESCENCE OF SMALLER BUBBLES • RESOLUTION FROM THE NEXT LARGER INTRAGRANULAR GAS BUBBLES • RELEASE FROM CRYSTALLINE DEFECTS	-	LOSS BY • DIFFUSION OUT OF THE GRAIN • GAS ATOM CAPTURE AND BUBBLE COALESCENCE • RESOLUTION • CRYSTALLINE DEFECT CAPTURE • GRAIN GROWTH	
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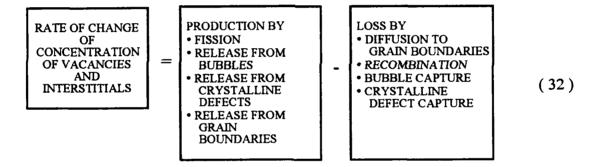
(30)

The same arguments, as for single gas atoms, can be used to eliminate effects from crystalline defects and grain growth. The bubble rate equation can thus be simplified:

RATE OF CHANGE
OF
OF
CONCENTRATION
OF GAS BUBBLES
OF 'N' ATOMS

$$= \begin{bmatrix}
PRODUCTION BY \\
• FISSION \\
• NUCLEATION OR
COALESCENCE OF
SMALLER BUBBLES
• RESOLUTION FROM
THE NEXT LARGER
INTRAGRANULAR
GAS BUBBLES
$$= \begin{bmatrix}
COSS BY \\
• DIFFUSION OUT OF
THE GRAIN
• GAS ATOM CAPTURE
AND BUBBLE
COALESCENCE
• RESOLUTION (COALESCENCE)
• RESOLUTION (COAL$$$$

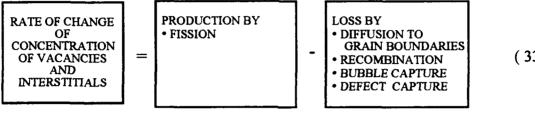
Vacancies and interstitials can be treated likewise with an overall equation:



Unlike gas atoms and bubbles, effects from crystalline defects are included as they provide the driving force for bubble growth [5,86]. Matthews and Wood [15] note that the effect of crystalline defects as vacancy sources will be negligible compared to the grain boundary for grains less than 100 μ m. However, vacancy effects from crystalline effects are important for determining the vacancy/interstitial balance, and thus vacancy emission from defects is included. Release of interstitials from sinks are considered negligible since the equilibrium concentration of interstitials in UO₂, and presumably UN, is negligible [5,87]. Release of vacancies from bubbles and

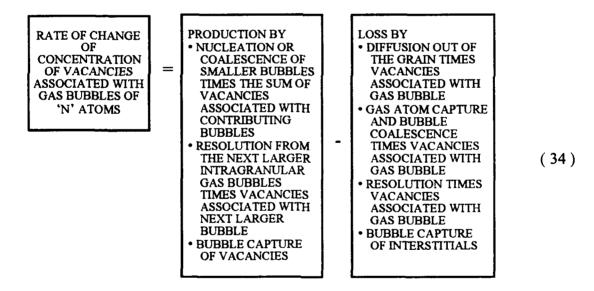
)

grain boundaries are not incorporated explicitly, but as boundary conditions when solving equations for the overall concentrations. Thus, the simplified equation for vacancies and interstitials is:



(33)

The rate equation describing bubble vacancy adjustment is a modified version of the simplified bubble rate equation:



The actual number of vacancies in a single bubble is obtained by dividing the solution to the above equation by the bubble concentration.

The form of the equation used in REDSTONE is the calculation of bubble radii, not the calculation of bubble vacancies. The bubble radius depends directly on the number of vacancies. The radius equation can thus be related to the above

equation describing bubble vacancies. Many of the properties discussed in earlier chapters also depend on the bubble radius. The bubble radius, in turn, indirectly depends on the fuel properties. Calculating the bubble radius directly allows one to update these properties concurrently, improving the stability of calculation. The specifics of converting from the bubble vacancy equation to the radius equation will be covered in later sections and chapters.

4.2 1-D Diffusion Equation in Spherical Coordinates

The diffusion equation for a single species (n=1 for atoms, n>1 for bubbles,n=v,i for vacancies, interstitials respectively) can be written in general terms as

$$\frac{\partial C_n}{\partial t} = \nabla \cdot \left(D_n \nabla C_n \right) + \alpha_n + \beta_n C_n + \gamma_n C_n^2 + \sum_{\substack{k=1\\k=n}}^N \rho_{nk} C_n C_k + \sum_{\substack{k=1\\k=n}}^N \eta_k C_k + \sum_{\substack{k=1\\k=n \ j=n}}^N \varepsilon_{kj} C_k C_j$$
(35)

where C_n = the unknown quantity of n'th species (concentration or vacancies per bubble), D_n = diffusion coefficient, α_n = constant factors, β_n = constant multipliers of the unknown quantity, γ_n = constant multipliers of the square of the unknown quantity, ρ_{nk} = constant multipliers of products of the unknown quantity and k'th species, ε_{kj} = constant multipliers of products of k'th and j'th species, η_k = constant multipliers involving the k'th species. Note that the 'j' index in the double summation term begins at 'k' rather than '1'. Since there is no distinction between contributions from ' $C_k C_j$ ' or ' $C_j C_k$ ', starting the summation at '1' would wrongly include a second contribution from this term. For 1-D spherical coordinates the gradient term, which describes the

diffusion loss out of the grain, becomes

$$\nabla \cdot (D_{n} \nabla C_{n}) = \frac{1}{r^{2}} \frac{\partial}{\partial r} \left(D_{n} r^{2} \frac{\partial C_{n}}{\partial r} \right) .$$
(36)

Since the diffusion coefficient is held constant during calculations, this equation becomes

$$\nabla \cdot \left(D_{\mathbf{n}} \nabla C_{\mathbf{n}} \right) = \frac{D_{\mathbf{n}}}{r^2} \frac{\partial}{\partial r} \left(r^2 \frac{\partial C_n}{\partial r} \right) . \tag{37}$$

The specific multipliers of the overall equation are determined by the individual processes. These processes are listed in table 4 by species type and which multiplier is affected. Individual processes are discussed in more detail by equation type below.

4.3 Gas Atom and Bubble Concentrations

With the exception of biased diffusion out of the grain, the processes governing the change in gas atom and bubble concentrations are described by

$$\frac{\partial C_{n}}{\partial t} = D_{n} \nabla^{2} C_{n} - K_{n} C_{n} + K_{n-1} C_{n-1} - b_{n} C_{n} N_{n} + b_{n+1} C_{n+1} N_{n+1} - 32 \pi R_{n} D_{n} C_{n}^{2} - \sum_{\substack{k=n \ k \neq n}}^{N} \left[4 \pi (R_{n} + R_{k}) (D_{n} + D_{k}) - \pi (R_{n} + R_{k})^{2} |V_{n} - V_{k}| \right] C_{n} C_{k} + \sum_{\substack{k=n \ k \neq n}}^{n/2} \left[4 \pi (R_{n-k} + R_{k}) (D_{n-k} + D_{k}) - \pi (R_{n-k} + R_{k})^{2} |V_{n-k} - V_{k}| \right] C_{n-k} C_{k}$$
(38)

where each of the terms are in order of appearance: diffusion, source loss from producing the next larger bubble, source gain from the next smaller bubble, resolution loss to the next smaller bubble, resolution gain from the next larger bubble, self-coalescence loss, coalescence loss from random and biased effects, and

Table 4

Identification	of Processes
----------------	--------------

	Single Atoms	Bubbles	Vacancies and Interstitials	Bubble V ich icy Adjustn.ca
α	PTOTAL FISSION GAS PRODUCTION RATE		TOTAL VACANCY AND INTERSTITIAL PRODUCTION RATE EFFECT OF BUBBLE OVERPRESSURE ON VACANCY CAPTURE	
β	 FRACTION OF PRODUCTION ATOMS DEPOSITED IN EXISTING BUBBLES LOSS BY RANDOM DIFFUSION OUT OF THE GRAIN 	 PRODUCTION BY FISSION (PROBABILISTIC) LOSS BY RANDOM DIFFUSION OUT OF THE GRAIN 	• LOSS FROM CAPTURE BY BUBBLES AND DEFECTS	• LOSS DUE TO DIFFUSION, RESOLUTION RATE, AND COALESCENCE RATE OF BUBBLE
γ	LOSS BY NUCLEATION OF GAS BUBBLES	LOSS BY ISO- BUBBLE COALESCENCE		
ρ	• LOSS BY GAS BUBBLE CAPTURE	• LOSS BY GAS ATOM CAPTURE AND BUBBLE COALESCENCE	LOSS FROM RECOMBINATION	
ε		PRODUCTION BY NUCLEATION OR COALESCENCE OF SMALLER BUBBLES		• GAIN FROM COALESCENCE OF BUBBLES
η	• PRODUCTION BY RESOLUTION FROM INTRAGRANULAR GAS BUBBLES	PRODUCTION BY RESOLUTION FROM THE NEXT LARGER INTRAGRANULAR GAS BUBBLES		• GAIN FROM RESOLUTION OF BUBBLES

coalescence gain from random and biased effects. Each will be discussed in the following sections.

4.3.1 Production by Fission

The production of gas atoms by fission can be represented by

$$K = Y_{\rm FG}\dot{F} \tag{39}$$

where Y_{FG} = yield of fission gas atoms per fission and F = fission rate. Both the yield and fission rate are taken as constant within a time step and spatially. This latter constraint is consistent with the fast neutron flux expected in the reactor core.

The yield of fission gas atoms per fission was taken as 0.25 [5] from consideration of the stable isotopes of krypton and xenon.

Researchers generally have considered the fission gas source to contribute only to single gas atoms. However, scoping calculations revealed that if gas atoms are immobile, then the final density of atoms is too great to be realistically supported by the lattice. Because of the high density, there is a distinct possibility that deposition of gas atoms in latter stages of operation would have a finite probability of forming new bubbles or increasing the size of existing bubbles. Such a process can be modeled by considering the overall fission gas source to be divided up into separate components for single atoms and each bubble size. Apportionment is determined by calculating the ratio of nearest neighbor lattice sites for a particular bubble size to the total number of lattice sites available within a unit volume. Lattice sites, whether filled or unfilled, are assumed to always exist and no calculation is done to alter the dimensions of the fuel grain. Since the number of lattice sites remains constant, the total number of lattice sites in a unit volume can be calculated once at the beginning of calculations, i.e.

$$N_{\rm L} = \frac{1}{a_{\rm o}^3} \tag{40}$$

where a_0 is the lattice constant. The number of nearest neighbor lattice sites for a single gas atom in a face-centered-cubic (FCC) structure is 12 [5]. The number of nearest neighbor lattice sites for a bubble can be calculated from the surface area of the bubble and the (average) density of lattice sites on the surface. When multiplied by the bubble concentration, C_n , this gives the total nearest neighbor sites for a bubble size in a unit volume. The bubble surface area is

$$A_{\rm n} = 4\pi R_{\rm n}^2 \tag{41}$$

where R_n is the radius of bubble with 'n' atoms. In calculating the density of lattice sites on a bubble surface, most investigators consider only the (100) crystal direction. A more accurate density can be calculated by considering the three principal directions (100), (110), and (111) and taking an average. Lattice site surface densities for each direction are:

$$\frac{2}{a_{o}^{2}} \qquad \frac{2}{\sqrt{2}a_{o}^{2}} \qquad \frac{5}{\sqrt{3}a_{o}^{2}} \qquad (42)$$
(100) (110) (111)

The average of these three directions is 2.10032 a_o^{-2} . Combining with the bubble surface area, the number of nearest neighbor sites to a single bubble is given by Nearest Neighbors = 8.40128 $\pi R_n^2 a_o^{-2}$. (43) Multiplying by the bubble concentration and fission gas source, and dividing by the

total number of lattice sites, the source term for bubble one atom larger than 'n' is given by

$$K_{\rm n} = \beta_{\rm n} C_{\rm n} = 8.40128\pi R_{\rm n}^2 a_{\rm o} C_{\rm n} Y_{\rm FG} \dot{F}$$
 (44)

Correspondingly, the source for single atoms deposited in the grain is given by

$$K_{o} = \alpha_{1} - \sum_{n=1}^{N} \beta_{n} C_{n} = Y_{FG} \dot{F} - \sum_{n=1}^{N} 8.40128 \pi R_{n}^{2} a_{o} C_{n} Y_{FG} \dot{F} .$$
 (45)

4.3.2 Nucleation

Historically, nucleation of bubbles has been treated homogeneously with modifying terms added to account for heterogeneous effects [88, 89, 90]. Heterogeneous nucleation has periodically been explicitly applied in an effort to explain anomalies in observed trends [91]. Heterogeneous nucleation was utilized in the BUBL code [87]. However, this code also assumed gas release from the grain occurred entirely from bubble motion and bubble interactions occurred only at dislocations. Single gas atoms in the matrix only served to nucleate and increase the size of bubbles on dislocation lines. Robertson [92] makes the observation that applied heterogeneous nucleation is on a scale so fine that little difference exists between the two approaches in practice. This is supported by the calculations of Dollins [93] which indicated that a gas atom is several orders of magnitude more likely to meet another gas atom than a dislocation. Also, Warner and Nichols [94] note that the parameter-fitting used to adjust models to actual experimental data minimizes any large discrepancies between the two models. In REDSTONE, homogeneous nucleation is applied with the only heterogeneous effect being the probabilistic production discussed in the previous section. This assumption appears legitimate since fission damage will be lower in UN than in UO_2 and crystalline defect concentration will be negligible compared to the bubble density.

Typically, a stable nucleus of a bubble is considered to be a diatomic cluster of gas atoms [88]. All researched modeling efforts have assumed this, and this will be assumed to be the case here. The modeling of bubble nucleation will introduce a loss term into the single gas atom equation and a gain term into the equation describing diatomic gas bubbles. Gain of diatomic bubbles through nucleation is described by an equation given by

Gain Nucleation Rate = $k_{11}C_1^2$ (46) where C_1 = gas concentration and k_{11} = rate constant. The nucleation loss term for single gas atoms is

Loss Nucleation Rate = $-2k_{11}C_1^2$. (47) The equation is multiplied by two because two gas atoms are removed for every diatomic cluster formed.

Following a purely theoretical approach, Olander [5] derived the rate constant for single vacancy processes, k_{vv} , in terms of a combinatorial number and basic properties of the fuel and gas atom, i.e.

$$k_{\rm vv} = \frac{z_{\rm vv}\Omega D_{\rm v}}{a_{\rm o}^2} \tag{48}$$

where $D_v =$ diffusion coefficient of vacancies, $z_{vv} =$ a to-be-determined combinatorial number, Ω = volume per lattice site = $a_0^3/4$ for a FCC lattice structure, and a_0 = lattice parameter. The combinatorial number as calculated by Olander is based on the number of nearest neighbors, the species jump frequency, and the species concentration. For a FCC structure, the calculated combinatorial number is 84. In considering interactions between xenon atoms, Olander states that the combinatorial number is probably higher than an atomic value (such as for vacancies) because evidence suggests that the xenon atoms migrate as a complex involving the xenon atom and several vacancies. Researchers have estimated that a single gas atom is combined with two to four vacancies, with the most probable being a neutral trivacancy in UO, [95,96]. For UN, there have not been any studies to determine the optimum number of vacancies associated with a gas atom. Since a neutral set of vacancies seems most logical, a gas atom in UN will be assumed to consist of the atom and a neutral divacancy. Olander does not calculate the corresponding combinatorial number for a xenon-vacancy complex nor does he give an indication of how much larger this number should be than the atomic value. If the number of occupied nearest neighbors to the complex are the controlling factor, as indicated by Olander's analysis, then a divacancy complex in UN would have 18 versus 12 nearest neighbor sites. Based on this change alone, the combinatorial number becomes 126. However, this is an overly simple adjustment and there are

probably other mitigating factors, such as the effect of divacancies associated with the jumping atom as well as whether a gas atom jumping into a nearest neighbor to a vacancy constitutes nucleation, which might reduce the combinatorial number. Other researchers have successfully utilized a single vacancy mechanism (combinatorial number = 84) to describe xenon migration [88]. Because of the lack of evidence concerning xenon-vacancy complexes in UN and thus the calculation of a combinatorial number, a combinatorial of 84 will be assumed. In order to convert from the vacancy form of the rate constant to a rate constant for xenon, Olander states that in the case of an atomic species other than a vacancy, all one needs to do is replace the vacancy diffusion coefficient. Interestingly, had the appropriate atomic diffusion coefficient been used in Olander's development of the above equation the vacancy fraction, $x_{v_{r}}$ would have appeared in the denominator of the rate constant. This happens because of Olander's definition of the diffusion coefficient for an atomic species, i.e.

$$D_{\rm x} = a_{\rm o}^2 x_{\rm v} w \tag{49}$$

versus the definition for the vacancy diffusion coefficient, i.e.

$$D_{\rm v} = a_{\rm o}^2 w \tag{50}$$

where w = the jump frequency of the species. One possible reason for this discrepancy is that an atomic species interacting with another species (gas atom, dislocation, bubble, grain boundary) will essentially 'see' a vacancy fraction of one due to the vacancies associated with the secondary interacting species. Certainly, the vacancy fraction would be expected to be greater than in the bulk matrix. If a vacancy fraction of one is not the case, then assuming it to be so will give the

maximum rate of interaction. Making appropriate substitutions and notation changes, the rate equation becomes

$$k_{11} = \frac{z_{11}\Omega D_1}{a_0^2} \tag{51}$$

where D_1 = diffusion coefficient of single xenon atoms and $z_{11} = z_{vv} = 84$. The rate constant for nucleation is given by

$$k_{11} = 21a_0 D_1 \tag{52}$$

leaving a nucleation rate of

Nucleation Rate =
$$42a_0D_1C_1^2$$
 (53)

where k_{11} was multiplied by two to account for both gas atoms being mobile [5].

Most recent researchers have expressed the nucleation rate from a less theoretical approach, based on the work of Chandrasekhar [34] who developed rate equations for use in colloidal chemistry. Rest [17] expresses the nucleation term as Nucleation Rate = $16\pi F_1 D_1 C_1^2$ (54) where F_1 = probability that two gas atoms which come together actually stick and R_1 =radius of volume occupied by the intragranular gas atom. Hayns and Wood [97] use a similar expression that does not include the sticking factor and is lower by

exactly a factor of two.

An effective radius of a gas atom, R_1 , can be calculated by assuming an atomic volume of $\Omega = a_0^3/4$ and equating this with the volume of a sphere, $\frac{4}{3}\pi R_1^3$, giving

$$R_{\rm I} = a_{\rm o} \sqrt[3]{\frac{3}{16\pi}} = 0.69336a_{\rm o} \ . \tag{55}$$

This gives a reaction rate (with $F_1=1$) based on Chandresakhar's work of Nucleation Rate = $34.8521a_0D_1C_1^2$ (56) which is approximately 20% less than that of Olander (45% if a combinatorial number of 126 is used). Esteves [68] has compared Chandresakhar's equations to more exact (and complex) theoretical rate equations [98] and concluded that both are of comparable accuracy. Given the relative closeness of the two results, the degree of uncertainty currently present in fuel modeling, and the conclusions of the above study, it is reasonable to utilize Chandresakhar's equations for bubble nucleation. Since they are also more efficient and straightforward in fuel modeling codes, their use in REDSTONE seems more than justified. It will be seen later that this has advantages as bubble coalescence and atom capture can be treated similarly. A sticking factor is not used in program REDSTONE as its use by Rest appears to be as a fitting parameter to adjust his code to actual data.

The previous discussion of nucleation was based on theory involving random effects. In a temperature gradient, atoms and bubbles will experience an additional force pushing them in a single direction at a speed proportional to the temperature gradient and their size. For nucleation, no additional biased terms are needed since all gas atoms at a particular location are pushed at the same velocity. Only random effects will contribute to the nucleation of bubbles.

4.3.3 Coalescence and Capture

Bubble coalescence and gas atom capture are expressed, as for nucleation, through a rate equation. In this case two concentrations must be considered, one for the bubble and one for either the gas atom or other bubble, i.e.

Coalescence Rate = $k_{kn}C_kC_n$ (57) where C_k, C_n = concentrations of bubbles with 'k' and 'n' atoms respectively and k_{kn} = rate constant for the reaction. Again, analysis of this equation will be

approached first from theoretical considerations and then from the equations of Chandrasekhar, the form typically used.

Theoretical analysis of the rate constant has been performed by Olander [5]. The method of analysis he uses depends on the predominant process controlling the reaction. Coalescence of reacting species (bubbles, atoms, or vacancies) which are of comparable size, and thus comparable mobilities, is considered to be reaction rate controlled, as was the nucleation rate developed in the previous section. When one of the species is much larger than the other, the process is diffusion limited due to the establishment of a concentration gradient of the more mobile species around the larger species. Between these two extremes a combination of these two processes exists. Interactions between gas atoms and bubbles will be treated as reaction rate controlled as bubble sizes are expected to be small in the scenario here. Interactions between bubbles will be treated likewise. Interactions between bubbles and vacancies/interstitials are treated as diffusion controlled, both theoretically and in practice, as vacancies and interstitials will have much higher diffusivities than the bubbles involved in the interaction. Diffusion controlled reactions will be discussed in the section dealing with vacancy-bubble interaction.

In general, bubble interaction rates based on reaction rate control are cumbersome to calculate from purely theoretical concerns. Partly, this is because one must calculate analogous terms to the jump distance and jump frequency used for atomic species. The difficulty in calculating these terms involves the more general question of what constitutes the formation of the coalesced bubble. Although the time required for a single coalescence is usually considered to be negligible, it is not instantaneous and involves the movement of many atoms from start to finish. This complication manifests itself in the calculation of the

combinatorial number, z_{kn} , used in the rate constant. However, the interaction of a single gas atom (treated as an atomic species) and a stationary bubble (gas atom capture) is simple enough that a combinatorial number can be calculated using a modification of Olander's analysis of vacancy-vacancy interactions. In the following analysis gas atom capture by a bubble will be considered, with the modification that the vacancy fraction will be taken as one.

The rate constant for gas atom capture, analogous to that for nucleation, is given by

$$k_{1n} = \frac{z_{1n} \Omega D_1}{a_0^2}$$
 (58)

where Ω = atomic volume, a_0 = lattice constant, z_{1n} = combinatorial number, and D_1 = diffusion coefficient of the single atom. Because only D_1 is present in the above equation, it is implicitly assumed that the bubble is stationary. In order to calculate a rate constant for two mobile species, it is necessary to calculate k_{n1} and add this to k_{1n} . It is for this reason that the nucleation rate constant in the previous section was multiplied by a factor of two $(k_{11}+k_{11})$. As discussed previously, the calculation of k_{n1} is complex as it requires the probability of a bubble 'jumping' next to a gas atom. The bubble has been assumed stationary for this reason. However, by considering only k_{1n} , an acceptable value for comparison can still be obtained since the magnitude of k_{n1} will be smaller than k_{1n} due to the gas atom being the more mobile species.

The development of the rate constant, following Olander, begins by first considering the rate of formation of bubbles containing 'n+1' atoms through capture of a single gas atom, i.e.

Interaction Rate =
$$P_{n+1}C_1$$

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(59)

where P_{n+1} = probability that a gas atom jumps into a site nearest neighbor to the bubble. A gas atom that jumps into a nearest neighbor site is assumed to become part of the bubble. The probability P_{n+1} depends on the number of nearest neighbors as well as the probability per second that a gas atom jumps into a particular site. The number of nearest neighbors to a bubble has previously been calculated to be Nearest Neighbors = $8.40128\pi R_n^2 a_o^{-2}$. (60)

Thus, the probability per second that a gas atom jumps into a nearest neighbor site is given by

$$P_{n+1} = 8.40128\pi R_n^2 a_o^{-2} P_x \tag{61}$$

where P_x = probability per second that a gas atom jumps into a particular site. The probability P_x can be obtained by considering the number of sites next to a specific nearest neighbor site which are not themselves nearest neighbors. One must also consider the probability that a particular site is occupied by a gas atom and the jump frequency of the gas atom in a particular direction.

The number of sites next to a nearest neighbor will range between four and seven. Four sites are present when considering a planar bubble surface (as for a large bubble), while seven sites are present in the case of nucleation. The probability that a particular site is occupied by a gas atom is given by $C_1\Omega$. The jump frequency of gas atoms in the matrix can be obtained from theory for the diffusion coefficient ($x_v = 1$), i.e.

$$D_1 = a_0^2 w \tag{62}$$

where w = jump frequency of xenon atoms. Rearranging, one gets the jump frequency in terms of measurable quantities,

$$w = \frac{D_1}{a_o^2}$$
 (63)

Putting these quantities together gives

$$P_{n+1} = \left(8.40128\pi \frac{R_n^2}{a_o^2}\right) \left(4C_1 \Omega\right) \left(\frac{D_1}{a_o^4}\right) = 33.6051\pi \frac{R_n^2}{a_o^4} D_1 C_1 \Omega \quad (64)$$

The rate of gas atom capture for bubbles of 'n' atoms is then

Capture Rate = 33.6051
$$\pi \frac{R_n^2}{a_o^4} D_1 C_1 \Omega C_n$$
 (65)

giving a rate constant of

$$k_{1n} = \frac{z_{1n}\Omega D_1}{a_o^2} = 33.6051\pi \frac{R_n^2}{a_o^4} D_1\Omega$$
 (66)

and a combinatorial number of

$$z_{1n} = 33.6051\pi \frac{R_n^2}{a_o^2} . \tag{67}$$

The value for a_0 for UN is approximately 4.889×10^8 cm. Substituting this into the rate equation gives

Capture Rate =
$$1.7184 \times 10^8 \pi R_n^2 D_1 C_1 C_n$$
 (68)

As with nucleation, many investigators [17,97] handle random gas atom capture following Chandrasekhar [34], i.e.

Rate =
$$4\pi (D_1 + D_n)(R_1 + R_n)C_1C_n$$
 (69)

The form of this equation is identical to that used for bubble coalescence, and similar to that for bubble nucleation. It thus provides a uniform theoretical treatment of gas atom and bubble interactions. Both bubble and gas atom motion are included, i.* neither species is assumed stationary. If the bubble is assumed stationary $(D_n=0)$ the rate equation becomes

Rate =
$$4\pi D_{\rm l}(R_{\rm l} + R_{\rm n})C_{\rm l}C_{\rm n}$$
 (70)

This equation differs from that derived using Olander's approach through a linear versus quadratic dependence on bubble radius. An inspection of the equation based on Chandrasekhar reveals that as the bubble becomes much larger than a single atom, the theoretical form of the diffusion limited reaction is achieved, i.e. Diffusion Limited Rate = $4\pi R_n D_1 C_1 C_n$. (71)

The equations of Chandrasekhar, at first glance, appear vastly different from the reaction limited equations based on a purely theoretical treatment (with its associated assumptions). A more direct comparison can be made by first considering nucleation and substituting for R_1 and R_n , which are both approximately 3.072×10^{-8} cm. In this case, the theoretical equation (with D_n assumed to be zero) becomes

Nucleation Rate =
$$1.62 \times 10^{-7} \pi D_1 C_1 C_n$$
 (72)

while the equation according to Chandrasekhar becomes

Nucleation Rate =
$$2.46 \times 10^{-7} \pi D_1 C_1 C_n$$
 (73)

The two rates are comparable and certainly within any error associated with the assumptions made. If, instead, a bubble of 25×10^{-8} cm radius is assumed for R_n , then the rates become

Theoretical Rate =
$$1.07 \times 10^{-5} \pi D_1 C_1 C_n$$
 (74)

and

Chandresakhar Rate =
$$1.12 \times 10^{-6} \pi D_1 C_1 C_n$$
 (75)

The difference between rates is now an order of magnitude and increasing as bubble size increases. Another shortcoming of the theoretical analysis was that the bubble was assumed to be stationary, at least relative to the gas atom, which is certainly not the case for analysis just presented. Thus the accuracy of the theoretical treatment as applied is questionable, even for the regime where reaction rate control would be dominate. It should be apparent from the above analysis that the theoretical approaches for bubble and gas atom interactions are not applicable over the full range of bubble sizes. The approach using the equations of Chandrasekhar does seem to accommodate the full range, with acceptable agreement for reaction rate control and convergence to the theoretical limit for diffusion control. The conclusion can be made that the theoretical approach as applied by Olander is inaccurate for the full range of bubble sizes and too difficult to use. The approach taken from the work of Chandrasekhar provides a viable alternative to describe interactions over the entire range of bubble sizes. The reaction rate for interaction of two bubbles, according to Chandrasekhar, would be

Rate =
$$4\pi (D_k + D_n)(R_k + R_n)C_kC_n$$
 (76)

. .

which is also the same form resorted to by Olander in applying coalescence to his own program and is the form chosen for use in REDSTONE. For self-coalescence loss, this rate is multiplied by two since two species of 'n' size are removed for each reaction.

For biased effects due to a temperature gradient, additional terms are required for any two species of different size. Bubbles of the same size would not interact for the same reason there is no biased nucleation. The force experienced by a bubble in a temperature gradient is due to the difference in surface atom energies between the 'hot' and 'cold' sides of the bubble. The atoms on the hot side will have more energy allowing more atoms to escape from the surface and travel along the inside of the bubble. Some of these atoms will come to rest on the cold side where they will lose their energy and remain. The result will be a net flux of surface atoms to the cold side of the bubble and a migration of the bubble up the

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temperature gradient. Biased bubble motion is treated as if a force is acting on the bubble as a whole, and is proportional to the bubble size and the temperature gradient within the bubble. Because of the lower thermal conductivity of the gas within the bubble, the temperature gradient within the bubble is higher than in the bulk matrix. For a spherical bubble, the effective temperature gradient on the bubble, ∇T_{br} is related to the temperature gradient in the bulk matrix, ∇T , by

$$\nabla T_{\rm b} = \frac{3}{2} \nabla T \ . \tag{77}$$

The velocity of a bubble in a temperature gradient is given by the Nernst-Einstein equation

$$V_{\rm n} = \frac{D_{\rm s} F_{\rm n}}{kT} \tag{78}$$

where k = Boltzman constant, $D_s = \text{Surface diffusion coefficient}$, and the force, F_n , is given by

$$F_{\rm n} = \left(\frac{2\pi R_{\rm n}^3}{a_{\rm o}^3}\right) \frac{Q_{\rm s}^*}{T} \nabla T_{\rm b} \tag{79}$$

where Q_s^* is the activation energy for surface atoms breaking loose from the surface. Although D_s has been measured for UN, Q_s^* has not. However, Olander [5] notes that the value recommended by Maiya [79] for UO₂ is approximately 80% of the heat of vaporization. This result is reasonable considering that jumps of surface atoms involve atoms which almost leave the surface permanently. Thus, an estimate of Q_s^* for UN can be obtained by assuming it to be 80% the heat of vaporization . For UN, this value has been reported as 1.1250×10^{-11} ergs [99], corresponding to a Q_s^* of 9.0003×10^{-12} ergs. In a given temperature gradient, it is the size of the bubble that determines the velocity. The coalescence rate will therefore depend directly on the relative velocities of the two bubbles, as well as the cross section for interaction, i.e.

Biased Coalescence Rate =
$$\pi (R_k + R_n)^2 |V_n - V_k| C_k C_n$$
 (80)

4.3.4 Resolution

The rate of resolution of gas atoms from bubbles is commonly treated by applying a resolution parameter, b, to the bubble concentration, i.e.

Resolution Rate =
$$bC_n N_n$$
 (81)

where C_n = concentration and N_n = number of gas atoms in the bubble.

Experimental measurements [100,101] have placed the resolution parameter for UO_2 in the range $1.8 \times 10^{-4} < |b| < 3.6 \times 10^{-4}$. Theoretically, the resolution parameter has been described by a variety of different methods based on various theories. One theory of resolution [102] attributes the process to the production of a thermal spike from a passing fission fragment in the vicinity of a gas bubble. The thermal spike creates temporary disorder in the lattice surrounding the bubble completely mixing fission gas atoms with matrix atoms. The simplest model of this theory [103] proposes that all bubbles intersected by a fission fragment track are entirely destroyed. Obviously, this model works best for small bubbles. The resolution parameter in this case is given by

$$b = 2\pi R_{\rm n}^2 \mu_{\rm ff} \dot{F} \tag{82}$$

where R_n = radius of bubble with 'n' atoms, μ_{ff} = distance travelled by a fission fragment during slowing down from birth energy, and $2\pi R_n^2$ = the bubble cross section for annihilation. Because whole bubbles are destroyed in the above theory, the loss rate of bubbles due to resolution is given by bC_n rather than bC_nN_n . A modification of the above model [104] suggests that the cross section for annihilation is not the bubble cross section, but a cross section determined by a cylinder of disorder created along the fission fragment track. The cross section for annihilation would then become $2\pi(R_n^2 - R_f^2)$, where R_f is the radius of the damage cylinder. Complete annihilation of a bubble would occur if it was totally within this cylinder. For bubbles larger than this cross section, a passing fission fragment would simply decrease the size of the bubble.

Another proposed theory of resolution [105] is that a passing fission fragment blasts off chunks of the matrix onto the opposite wall of a bubble through a pressure pulse, trapping fission gas in the process. Full conditions for the resolution processes above have been discussed by Blank and Matzke [8]. However, the above theories probably do not describe resolution in a fuel like UN, which has a higher thermal and electrical conductivity than UO_2 [8,106]. For UN the resolution constant is expected to be at least an order of magnitude less.

Other resolution models [10,107,108], based on theory attributed to Nelson, propose a knock-on process whereby the fission fragment physically knocks individual gas atoms out of the gas bubble into the matrix. In this theory, the resolution parameter is given by [5]

$$b = \int_{E_{\rm ff}^{\rm max}} \phi(E_{\rm ff}) \sigma(E_{\rm ff}, T) dE_{\rm ff} = 2\sigma(E_{\rm ff}^{\rm max}, T_{\rm min}) \ln\left(\frac{E_{\rm ff}^{\rm max}}{T_{\rm min}}\right) \mu_{\rm ff} \dot{F}$$
(83)

where $E_{\rm ff}$ = the energy of a fission fragment at some point in the slowing down process, $E_{\rm ff}^{\rm max}$ = the maximum energy of a fission fragment, $\phi(E_{\rm ff})$ = the energy spectrum of the fission fragment flux, $T_{\rm min}$ = minimum energy required to completely redissolve the gas atom, $\sigma(E_{\rm ff},T)$ = transfer cross section for energy T to T+dT from a fragment of energy $E_{\rm ff}$. This resolution parameter is highly dependent on the minimum energy needed for resolution. This minimum energy will increase with increasing temperature since gas atom mobility at higher temperatures will make it easier for an escaped gas atom to diffuse back to the bubble. Olander calculates the above resolution parameter using values of $E_{\rm ff}^{\rm max}$ =67MeV, $\mu_{\rm ff}$ = 6 micrometers, and two values of $T_{\rm min}$, 300 eV and 1 keV. The calculated values are $b=1.1\times10^{-19}F$ and $b=4.1\times10^{-19}F$ for $T_{min}=300$ eV and 1 keV, respectively. These calculated values are about an order of magnitude lower than experimental values in UO2. Nelson [10] tried to correct for this by estimating the additional effect of resolution from secondary knock-ons. However, in a fuel such as UN, it is not clear that secondary knock-ons would be produced as easily as in UO₂. For UC, the theoretical single knock-on resolution rate has been shown to be a satisfactory estimate of the experimental resolution rate [9]. Since UN has similar electrical properties, the same conclusion is made for UN. Thus, values of the resolution parameter used in REDSTONE come from the theoretical calculated values of Olander. Resolution at low temperatures (<500 K) utilizes the T_{\min} =300eV value, while resolution at high temperatures (> 1500 K) utilizes the $T_{\min}=1$ keV value, with prorated values between these two extremes.

The above resolution parameter is all that is required for small bubbles. For large bubbles, the energy imparted to a gas atom in the center of the bubble may not be enough to force the atom past other gas atoms and out of the bubble. Only gas atoms located within a distance, d, of the surface will have enough energy to be ejected from the bubble. Nelson estimates this distance to be about 15 Å, increasing with decreasing bubble gas density according to the relationship,

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$$d = \frac{1/B}{\rho_{\rm n}} \tag{84}$$

where B = the van der Waals constant for xenon and ρ_n = the gas bubble density. From the van der Waals equation of state and the proportion of the outer shell of thickness, d, to the total bubble volume, a resolution efficiency for large bubbles is obtained, i.e.

$$\eta_{n} = 1 - \left\{ 1 - 15 \left[\frac{1}{R_{n}} + \frac{1}{B} \left(\frac{kT}{2\gamma} \right) \right] \right\}^{3}$$
(85)

where R_n = the bubble radius (Å), B = the van der Waals constant, k = Boltzman constant, and γ = surface energy. Thus the resolution rate for bubbles with R_n >15Å is

Resolution Rate =
$$\eta_n b C_n N_n$$
 (86)

Since resolution introduces single gas atoms back into the matrix, the summation of all resolutions of bubbles provides a source term for single gas atoms.

4.3.5 Diffusion Loss Due to a Temperature Gradient

When the grain is subjected to a temperature gradient, gas atoms and bubbles will be forced in a single direction without regard to the symmetry of the grain. For long term operation, this introduces serious complications to the modeling of the fuel grain, i.e. one must develop additional equations and methods to handle the 3-D problem. Because of the difficulty of modeling the 3-D equations, unique solving methods are required. Fortunately, the scenario to be modeled here introduces few biased effects as a result of 1) the essentially negligible temperature gradient during long term operation, and 2) the short duration of operation when a temperature gradient becomes important. The simplifications to the 3-D case to be made here are designed to retain as much of the biased effects as possible, but still maintain 1-D symmetry in the grain. Biased effects have already been introduced into the coalescence terms. In this section, a method will be described for accounting for gas and bubble loss to the grain boundary due to the temperature gradient.

In relation to a spherical fuel grain, the effect of a biased force will be to shift all bubbles of the same size a distance 'h' of a time period Δt . This distance is determined by the velocity of the bubbles. In the case of different sized bubbles, a separate 'h' would be associated with each size. Other investigators [15,109,110] had tackled this asymmetry problem by simply calculating the amount of overlap between two identical spheres separated by a distance 'h'. The fraction of this volume to the total volume of one of the spheres was taken to be the fraction of fission gas retained in the fuel grain. This appears to be the only method used by 1-D codes to model transient release of fission gas from the grain. The volume of the overlapping regions is given by [111]

$$V = \frac{\pi h^2}{6} \left(3R_{\rm g} - \frac{h}{2} \right)$$
 (87)

where R_g = grain radius. Dividing this by the volume of the sphere, $\frac{4}{3}\pi R_g^3$, and subtracting from one gives the fraction released, i.e.

Release Fraction =
$$1 - \frac{h^2(6R_g - h)}{16R_g^3}$$
. (88)

The distance 'h' is determined by multiplying the bubble velocity by the time step. The bubble velocity used is the average across the fuel grain for a particular bubble size. A sphere the size of the fuel grain is then shifted to determine the percentage loss of bubbles of that size. The percentage loss is applied to all nodes in the grain. Each bubble size is considered separately, but elements and nodes are not. This relatively simplistic approach was used because of the difficulty of accounting for asymmetric movement of bubbles from element to element and node to node.

4.4 Bulk Vacancy and Interstitial Concentrations

The processes governing bulk vacancy change in the fuel grain are described by

$$\frac{\partial C_{\rm v}}{\partial t} = D_{\rm v} \nabla^2 C_{\rm v} + K_{\rm v} - \alpha C_{\rm v} C_{\rm i} - Z_{\rm v} D_{\rm v} \rho_{\rm d} C_{\rm v}$$
$$- \sum_{k=1}^{N} 4\pi R_{\rm n} D_{\rm v} C_{\rm n} \left\{ C_{\rm v} - C_{\rm v}^{\rm e} \exp\left[-\frac{\Omega}{kT} \left(P_{\rm n} - \frac{2\gamma}{R_{\rm n}} - P_{\rm o} \right) \right] \right\}$$
(89)

where the terms are in order of appearance: diffusion, source, loss to dislocations, and loss to bubbles. The corresponding equation for interstitials is

$$\frac{\partial C_i}{\partial t} = D_i \nabla^2 C_i + K_i - \alpha C_v C_i - Z_i D_i \rho_d C_i - \sum_{k=1}^N 4\pi R_n D_i C_n C_i$$
(90)

where each of the terms are analogous to those in the equation for vacancies. Effects from grain boundaries are accounted for through appropriate boundary conditions, not as explicit terms in the equations. Each of the above terms will be discussed in the following sections.

4.4.1 Production by Fission

The production of vacancies and interstitials via fission has not been experimentally investigated for UN. For UO₂, researchers have estimated a production rate of $10^4 F \Omega$ to $5 \times 10^5 F \Omega$ [112, 113] with better agreement to experiment being obtained for values close to $5 \times 10^5 F \Omega$. Since production of fission damage in UN is expected to be lower than in UO₂, one would expect that production of vacancies and interstitials would also be lower. An estimate of the production rate in UN can be obtained by comparing experimental values for fission-enhanced diffusion in both UN and UO₂. Since fission-enhanced diffusion is attributed to the vacancies produced from fission, such a comparison should provide a basis to estimate vacancy production in UN from the corresponding value for UO₂. Fission-enhanced diffusion was discussed in section 2.3.2.2. In that section, it was seen that the diffusion rate in UO₂ was one to two orders of magnitude larger than that in UN. Assuming the vacancy and interstitial production rates in UN are the same order of magnitude lower, the production rates for UN can be estimated to be $K_v = K_i = 5 \times 10^3 \dot{F}\Omega$. (91)

4.4.2 Recombination

Olander [5] has given a detailed discussion of the interactions between point defects, expressing rate constants in terms of combinatorial numbers. His approach, discussed in previous sections, assumed that defects must jump into nearest neighbor positions to interact. However, Gibson [114] has shown that vacancies have a large sphere of influence extending beyond simple nearest neighbors. Olander does not specify what the actual sphere of influence for recombination is, except to suggest that the combinatorial number is approximately 10 times larger (approximately 100 atom volumes) than the nearest neighbor treatment (12 atom volumes). Straalsund and Guthrie [67] use a sphere of influence ranging between 12 and 500 atom volumes. Dollins and Jurisch [71] use a combinatorial number of 5040, equivalent to an atom volume of about 500 atoms. Using this number, the loss rate due to recombination is

Recombination Rate =
$$\alpha C_v C_i = \frac{5040 \Omega (D_i + D_v)}{a_o^2} C_v C_i$$
 (92)

where a_0 is the atomic jump distance. Although it is clear that a simple nearest neighbor treatment is insufficient for recombination, the exact sphere of influence to use is unclear. For program REDSTONE, the loss rate utilized by Dollins and Jurisch is adopted since it does provide for a larger sphere of influence.

4.4.3 Dislocation Interaction

Although dislocations are assumed to have a negligible effect on bubbles, they are included for vacancies and interstitials because studies have indicated that the slight bias dislocations have for interstitials provides the driving force for void growth [5,67,115]. One of the difficulties in including dislocations is determining their types and densities. To complicate matters, both of these can change over time with dose and temperature. Theoretical treatments have attempted to account for different dislocation types and applied them to the behavior in structural materials such as steel [115,116]. However, dislocation properties are not well documented for nuclear fuels, especially under irradiation conditions. Thus, for nuclear fuels, the applied dislocation treatment has been much more generalized, with dose and temperature effects not included. In fact, Rest [117] leaves out dislocations completely. Typically, the interaction terms are described by

Vacancy - Dislocation Interaction Rate = $Z_v \rho_d D_v \left[C_v - C_v^e \exp\left(\frac{\Omega}{kT} P_o\right) \right]$ (93) and

Interstitial - Dislocation Interaction Rate = $Z_i \rho_d D_i C_i$ (94) where thermal emission of interstitials is considered negligible and Z_v , Z_i = numbers characterizing capture volumes associated with unit length, ρ_d = overall dislocation density, C_v , C_i = concentrations, D_v , D_i = diffusion coefficients, and C_v^e = thermal

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(equilibrium) concentration. Because a good correlational predictor of dislocation types and densities for nuclear fuels is not available, this same set of equations is used in REDSTONE. The obvious limitations of this approach are that the dislocation density is held constant and only one type of dislocation is considered.

The numbers Z_v and Z_i are typically on the order of unity. However, because of the larger strain field surrounding an interstitial, Z_i is greater than Z_v by one to two percent. This difference is provides the driving force for steady-state void growth. Consistent with other researchers, the values of Z_v and Z_i in REDSTONE are taken to be 1.00 and 1.02 respectively.

The dislocation density was adopted from other researchers as well. Values of 10^8 cm/cm³ to 10^{11} cm/cm³ have been used [67,118,119]. Since UN is expected to have fewer dislocations than other corresponding materials in an irradiation environment, the lower value of 10^8 cm/cm³ is used in REDSTONE.

4.4.4 Bubble Interaction

Vacancy and interstitial diffusion rates are much greater than corresponding bubble diffusion rates. As a result, the rate at which vacancies and interstitials interact with a bubble can be calculated via diffusion equations in the vicinity of a single stationary bubble (diffusion limited reaction), given by

$$\frac{\partial C_{\rm v}}{\partial t} = \frac{D_{\rm v}}{r^2} \frac{\partial}{\partial r} \left(r^2 \frac{\partial C_{\rm v}}{\partial r} \right) + K_{\rm v} \tag{95}$$

and

$$\frac{\partial C_i}{\partial t} = \frac{D_i}{r^2} \frac{\partial}{\partial r} \left(r^2 \frac{\partial C_i}{\partial r} \right) + K_i \tag{96}$$

with the boundary conditions

$$C_{\rm v} = C_{\rm v}^{\rm e} \exp\left[-\frac{\Omega}{kT}\left(P_{\rm n} - \frac{2\gamma}{R_{\rm n}} - P_{\rm o}\right)\right] \quad \left\{ r = R_{\rm n} \right\}$$

$$C_{\rm i} = C_{\rm i}^{\rm e} = 0 \qquad (97)$$

and

$$C_{v} = C_{v}^{b}$$

$$C_{i} = C_{i}^{b}$$

$$\frac{dC_{v}}{dr} = 0$$

$$\frac{dC_{i}}{dr} = 0$$

$$r = R_{c}$$
(98)

plus initial conditions, where C_v, C_i = vacancy and interstitial concentrations,

 $D_{vn}D_i$ =vacancy and interstitial diffusion coefficients, $K_{vn}K_i$ = vacancy and interstitial production rates, R_c = capture volume associated with the bubble, and R_n = bubble radius. The effects from non-equilibrium bubbles have been included as boundary conditions for vacancies. A similar boundary equation would normally be present for interstitials except that the interstitial equilibrium concentration has been taken as zero. Capture effects from dislocations are not included in these equations as they are already included in the overall bulk concentration equations and bubbles are the dominant sink on vacancies [93]. Also, since it is desirable to come up with an analytic expression, recombination is not included here so that the equations remain uncoupled. Recombination is also included in the calculation of the bulk vacancy and interstitial concentrations. Solutions to the resulting equations are still not easily obtained in finite form because of the time derivative. Additionally, establishment of the initial condition for the time variable involves knowing the exact profile around each bubble size at the beginning of each time step. This requirement is not practical since it involves modeling vacancy profile changes for all bubbles. A greatly simplified solution can be obtained by taking a quasi-steady-state approach, i.e. setting the left hand side of the above equations to zero. Such an approach assumes that the vacancies and interstitials equilibriate around a bubble much faster than the bubble changes in size. This assumption is reasonable since one can argue that after the initial establishment of bubble profiles, all bubbles would have vacancy and interstitial profiles established around them. Subsequent coalescences, resolutions, etc would not change these profiles much from their equilibrium positions. Re-equilibriation of vacancies/interstitials would occur rapidly since only minor changes would be required. In support of this, Matthews [15] has concluded that a steady-state treatment of vacancies is appropriate in most situations. With this simplification, the equations become

$$0 = \frac{D_{\rm v}}{r^2} \frac{\mathrm{d}}{\mathrm{d}r} \left(r^2 \frac{\mathrm{d}C_{\rm v}}{\mathrm{d}r} \right) + K_{\rm v} \tag{99}$$

and

$$0 = \frac{D_i}{r^2} \frac{\mathrm{d}}{\mathrm{d}r} \left(r^2 \frac{\mathrm{d}C_i}{\mathrm{d}r} \right) + K_i \tag{100}$$

with boundary conditions as before and no initial conditions.

Olander notes that the solutions to these equations exhibit two distinct regions within the capture volume, each dominated by a different process. In the inner region a diffusion loss term is dominant and production effects can be neglected. In the outer region diffusion loss is negligible and only production need be considered. Thus approximate solutions in the inner region can be found by solving for the equations

$$\frac{1}{r^2}\frac{\mathrm{d}}{\mathrm{d}r}\left(r^2\frac{\mathrm{d}C_{\mathrm{v}}}{\mathrm{d}r}\right) = 0 \tag{101}$$

and

$$\frac{1}{r^2}\frac{\mathrm{d}}{\mathrm{d}r}\left(r^2\frac{\mathrm{d}C_i}{\mathrm{d}r}\right) = 0 \ . \tag{102}$$

The outer region is actually described by the overall production/loss equations describing bulk vacancy/interstitial concentrations. The concentrations change little in the outer region and quickly attain the values in the bulk matrix.

Usually, the capture radius, R_c , is taken to be much greater than the bubble radius, R_n . Assuming this to be true here, the flux boundary conditions at $r = R_c$ given earlier can be replaced with concentration boundary conditions at $r = \infty$, i.e.

$$C_{\mathbf{v}} = C_{\mathbf{v}}^{\mathbf{b}} \\ C_{\mathbf{i}} = C_{\mathbf{i}}^{\mathbf{b}}$$
 $r = \infty$ (103)

where a 'b' superscript has been added to denote concentrations far away from bubbles. These boundary conditions essentially convert the capture volume to an infinite medium. The solutions to these greatly simplified equations are,

$$C_{\mathbf{v}} = C_{\mathbf{v}}^{\mathbf{c}} + \left(C_{\mathbf{v}}^{\mathbf{b}} - C_{\mathbf{v}}^{\mathbf{c}}\right) \left(1 - \frac{R_{\mathbf{n}}}{r}\right)$$
(104)

and

$$C_{\rm i} = C_{\rm i}^{\rm b} \left(1 - \frac{R_{\rm n}}{r} \right) \,. \tag{105}$$

The resulting reaction rate for a single bubble is given by

Reaction Rate =
$$\left(-4\pi R_n^2\right) J$$
 (106)

where J is the flux across the bubble surface given by

$$J = -D\left(\frac{\mathrm{d}C}{\mathrm{d}r}\right)_{R_{\mathrm{n}}} \,. \tag{107}$$

Making the appropriate substitutions, the net single bubble interaction rates become

Vacancy Interaction Rate =
$$4\pi R_n D_v \left\{ C_v - C_v^c \exp\left[-\frac{\Omega}{kT} \left(P_n - \frac{2\gamma}{R_n} - P_o \right) \right] \right\}$$
 (108)

and

Interstitial Interaction Rate =
$$4\pi R_n D_i C_i$$
 (109)

where the 'b' superscript has been dropped from the bulk concentrations. To get the interaction rates within a unit volume, one needs to multiply by the concentration of bubbles in the volume, i.e.

Vacancy Interaction Rate =
$$4\pi R_n D_v C_n \left\{ C_v - C_v^e \exp\left[-\frac{\Omega}{kT} \left(P_n - \frac{2\gamma}{R_n} - P_o \right) \right] \right\}$$
 (110)

and

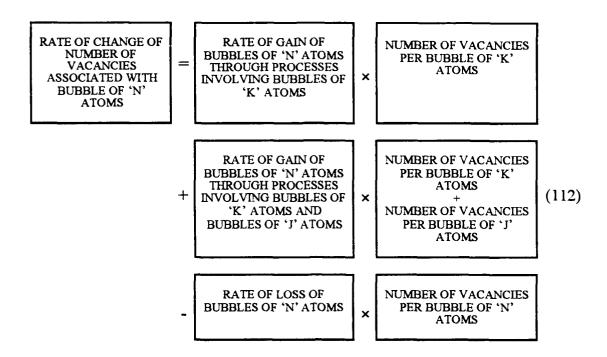
Interstitial Interaction Rate =
$$4\pi R_n D_i C_i C_n$$
 (111)
Previous investigators have used the forms above for use in transient modeling [71].

4.5 Bubble Radius Adjustment

A bubble vacancy adjustment equation was described in section 4.1 in terms of the various processes involved. In this section, this equation will be related to the concentration equation, and then the conversions needed to directly describe bubble radius changes will be discussed. Many of the relationships discussed here have already been discussed previously.

4.5.1 Relation to Bubble Concentration Equation

Ignoring for the moment the affect of the bulk vacancy concentration on the growth of fission gas bubbles, the resulting equation for bubble vacancy growth is related to the bubble concentration equations by noting



Stated in words, this conversion is simply the loss terms in the original concentration equation multiplied by the number of vacancies per bubble for the bubble being lost, plus the gain terms multiplied by the sum of the appropriate number of vacancies per bubble for the bubbles being gained. To complete the equation, one must add the effects from bubble-vacancy and bubble-interstitial interactions developed in section 4.4.4. This equation describes the rate of change of the total vacancy concentration associated with bubbles of 'n' atoms in a unit volume. The total vacancies associated with bubbles of 'n' atoms, C_{nv} , is equal to $C_n \times N_{nv}$, where C_n is the bubble concentration and N_{nv} is the number of vacancies per bubble of 'n' atoms. In REDSTONE, the bubble concentration is held constant during bubble volume adjustment calculations. The resulting equation with all terms included is

$$\frac{\partial C_{nv}}{\partial t} = \left(D_{n} \nabla^{2} C_{n} \right) N_{nv} - K_{n} C_{n} N_{nv} + K_{n-1} C_{n-1} N_{(n-1)v}
- b_{n} C_{n} N_{n} N_{nv} + b_{n+1} C_{n+1} N_{n+1} N_{(n+1)v} - 32 \pi R_{n} D_{n} C_{n}^{2} N_{nv}
- \sum_{\substack{k=n \ k \neq n}}^{N} \left[4 \pi (R_{n} + R_{k}) (D_{n} + D_{k}) - \pi (R_{n} + R_{k})^{2} |V_{n} - V_{k}| \right] C_{n} C_{k} N_{nv}
+ \sum_{\substack{k=n \ k \neq n}}^{N/2} \left[4 \pi (R_{n-k} + R_{k}) (D_{n-k} + D_{k}) - \pi (R_{n-k} + R_{k})^{2} |V_{n-k} - V_{k}| \right] C_{n-k} C_{k} (N_{kv} + N_{(n-k)v})
- 4 \pi R_{n} D_{i} C_{i} C_{n} + 4 \pi R_{n} D_{v} C_{n} \left\{ C_{v} - C_{v}^{*} \exp \left[-\frac{\Omega}{kT} \left(P_{n} - \frac{2\gamma}{R_{n}} - P_{o} \right) \right] \right\}$$
(113)

where each of the terms have been discussed previously.

4.5.2 Relation to Bubble Vacancy Equation

The bubble vacancy rate equation presented in the previous section is not the exact form used in REDSTONE. The equation in REDSTONE describes the rate of change of the radius of a bubble, which involves a conversion of the bubble vacancy equation. The conversion is made by noting that vacancies in a bubble are related to the radius by

$$N_{\rm nv} = \frac{4\pi R_{\rm n}^3}{\Omega} \tag{114}$$

where Ω is the atomic volume for one vacancy. Applying this conversion and rearranging somewhat, produces the following equation for radius change,

$$C_{n} \frac{\partial R_{n}}{\partial t} = \left(D_{n} \nabla^{2} C_{n} \right) \frac{R_{n}}{3} - K_{n} C_{n} \frac{R_{n}}{3} + K_{n-1} C_{n-1} \frac{R_{n-1}^{3}}{3R_{n}^{2}} - b_{n} C_{n} N_{n} \frac{R_{n}}{3} + b_{n+1} C_{n+1} N_{n+1} \frac{R_{n+1}^{3}}{3R_{n}^{2}} - 32\pi R_{n} D_{n} C_{n}^{2} \frac{R_{n}}{3} - \sum_{\substack{k=1\\k\neq n}}^{N} \left[4\pi (R_{n} + R_{k}) (D_{n} + D_{k}) - \pi (R_{n} + R_{k})^{2} |V_{n} - V_{k}| \right] C_{n} C_{k} \frac{R_{n}}{3} + \sum_{\substack{k=1\\k\neq n}}^{N/2} \left[4\pi (R_{n-k} + R_{k}) (D_{n-k} + D_{k}) - \pi (R_{n-k} + R_{k})^{2} |V_{n-k} - V_{k}| \right] C_{n-k} C_{k} \left(\frac{R_{k}^{3}}{3R_{n}^{2}} + \frac{R_{n-k}^{3}}{3R_{n}^{2}} \right) + \frac{Q}{R_{n}} D_{i} C_{i} C_{n} + \frac{Q}{R_{n}} D_{v} C_{n} \left\{ C_{v} - C_{v}^{e} \exp \left[-\frac{Q}{kT} \left(P_{n} - \frac{2\gamma}{R_{n}} - P_{o} \right) \right] \right\}.$$
(115)

In addition to this conversion, several of the properties depend on bubble radius. These include the diffusion coefficients, production rates, bubble pressures, and resolution rates. Each of these dependencies are reviewed in following sections. These should be considered in the formulation of the finite element problem. Unfortunately, inclusion of these adds even more non-linearity to the problem. Preliminary investigations indicated that only inclusion of bubble pressure dependencies made a noticeable difference in the convergence behavior. Additionally, calculational stability could be adversely affected by allowing the terms used to calculate concentrations to change when calculating radii. Fortunately, bubble pressure is only used in the vacancy interaction terms which were not used to calculate concentrations. Disregarding the other radius dependent property terms and keeping terms common with the concentration equation constant during the radius calculation simplifies the finite element formation of the radius equation considerably. The details of the formulation will be discussed in Chapter VII.

4.5.3 Radius Dependent Properties

Following is a short review of the radius dependence of fuel properties used in program REDSTONE.

4.5.3.1 Diffusion Coefficient

The diffusion coefficient has been defined previously in section 3.3.2.2 as

$$D_{\rm n}^{\rm new} = \frac{6\lambda\Omega D_{\rm s}}{\pi\lambda_{\rm s}^2 R_{\rm n}^2} \sin^2\left(\frac{\lambda_{\rm s}}{2R_{\rm n}}\right) \qquad \qquad R_{\rm n} > \frac{\lambda_{\rm s}}{\pi} \qquad (116)$$

$$D_{n}^{new} = \frac{6\lambda\Omega D_{s}}{\pi\lambda_{s}^{2}R_{n}^{2}} \qquad \qquad R_{n} \le \frac{\lambda_{s}}{\pi} \qquad (117)$$

where the included terms have been defined previously. The diffusion coefficient is seen to have an R_n^2 dependence modified for larger bubbles by the sin²() term which also depends on bubble radius.

4.5.3.2 Production Rate

The production rate of bubbles through deposition of new fission gas depends on the fraction of the bubble surface area and the total atomic sites, i.e.

$$K_{\rm n} = \frac{8.40128\pi\Omega}{a_{\rm o}^2} R_{\rm n}^2 \tag{118}$$

where the included terms have been defined previously. This equation indicates an R_n^2 dependence.

4.5.3.3 Bubble Pressure

The bubble pressure is described by the van der Waals reduced equation of state,

$$P_{\rm n} = \frac{N_{\rm n} kT}{\frac{4}{3} \pi R_{\rm n}^3 - N_{\rm n} B}$$
(119)

where the included terms have been defined previously. The bubble pressure is seen to have a more complex radius dependence. The equation is still in a form, however, that can be dealt with effectively.

4.5.3.4 Resolution Rate

For bubbles with radius greater than 15 Å the resolution parameter is described by

$$b_{\rm n} = b\eta_{\rm n} = b \left(1 - \left\{ 1 - 15 \left[\frac{1}{R_{\rm n}} + \frac{1}{B} \left(\frac{kT}{2\gamma} \right) \right] \right\}^3 \right)$$
(120)

where the included terms have been defined previously. For smaller bubbles, there is no radius dependence. The resolution parameter for larger bubbles is again seen to have a more complex radius dependence.

CHAPTER V CONCENTRATION FINITE ELEMENT FORMULATION

5.1 Generalities

Detailed modeling of the fission gas behavior within a fuel grain must take into account not only the individual gas atoms but also all possible gas bubble sizes. In reality, the number of bubble sizes modeled is either capped at some reasonable level or the bubbles are grouped into one or more 'average' bubble sizes which exhibit properties which are supposedly an average of the group. This latter approach has been the preferred method in recent years as it greatly reduces storage and running time requirements. Normally, the average group values are calibrated to agree with release results rather than swelling results, i.e. bubble group values are based on a bubble containing the average number of gas atoms in the group and bubble volume is not conserved. Programming the collapse of bubble sizes into groups is only done after multi-bubble size calculations are performed, confirming the collapsed results. This has not been done for UN. Since the study here concerns calculating both the release and the swelling, the decision was made to model individual bubble sizes rather than groups. Scoping calculations had indicated that the number of different bubble sizes formed would be small and therefore manageable within the storage and time constraints of the program. Consideration of the gas atoms as well as 'N' bubble groups results in 'N+1' non-linear equations that must be solved simultaneously. Because of the complexity and non-linearities of these equations, numerical methods must be used to solve them.

The diffusion equation can be expressed in the following general form

$$\frac{\partial C_{n}}{\partial t} = D_{n} \nabla^{2} C_{n} + \alpha_{n} + \beta_{n} C_{n} + \gamma_{n} C_{n}^{2} + \sum_{\substack{k=1\\k\neq n}}^{N} \rho_{nk} C_{n} C_{k} + \sum_{\substack{k=1\\k\neq n}}^{N} \eta_{k} C_{k} + \sum_{\substack{k=1\\k\neq n}}^{N} \sum_{\substack{j=k\\k\neq n \neq n}}^{N} \varepsilon_{kj} C_{k} C_{j}$$
(121)

This is the same equation presented in Chapter III except that the diffusion coefficient is constant here. This is also the form of the equation used in REDSTONE for bubble and vacancy/interstitial concentrations. The form used for bubble volume adjustment is somewhat different due to higher non-linearities and will be discussed in the next chapter.

5.2 Discretization of the Time Derivative

Note that in the general equation the concentration appears as a function of both time and position. A finite element formulation could be performed incorporating both these variables, but this would require nodal discretization in time as well as position. Such a discretization would require a single solution of a set of equations describing all concentrations at all nodal positions and at all discretized times. Such a formulation would be unmanageable for any reasonable problem. Instead, the time dependence is usually treated with finite difference techniques and finite elements are used for the remaining position variables. In the formulation of the current problem the partial derivative in time is replaced with a backward Euler approximation so that only a finite element formulation with position need be done.

A backward Euler approximation given by

$$\frac{t+\Delta t}{\partial t} \approx \frac{t+\Delta t}{\Delta t}$$
(122)

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was chosen as it does not impose a maximum time step size and thus is unconditionally stable. The notation used is that the unknown quantity is at time 't+ Δ t', and the last calculated known quantity is at time 't'. The unconditionally stable feature is important when dealing with non-linear problems because divergence of time stepping schemes is not always apparent. In other words, a non-linear problem using a central difference or forward Euler time stepping scheme may eventually converge even though the solution became unstable somewhere in the process. As a result, an answer can be obtained which is totally meaningless. Although a backward Euler scheme provides the stability needed in a non-linear problem, it has disadvantages verses both the other schemes.

Against a central difference scheme, i.e.

$$\frac{t+\Delta t}{\partial t} \approx \frac{t+2\Delta t}{2\Delta t}$$
(123)

which is second order accurate, the backward Euler scheme is only first order accurate. However, this trade-off is more than acceptable in ligh of the poor accuracy of experimental data in UN which must be input into the calculation. The advantage the forward Euler scheme,

$$\frac{t+\Delta t}{\partial t} \approx \frac{t+2\Delta t}{\Delta t} C_{n} - \frac{t+\Delta t}{\Delta t} C_{n}$$
(124)

over the backward Euler scheme is not accuracy, but has to do with the efficiency of the resulting program. The forward Euler scheme is an explicit method. As a result, for a linear problem time stepping involves only updates to the force vector The use of a forward Euler scheme allows one to calculate the stiffness matrix only once at the beginning of the problem because the non-time dependent stiffness terms remain constant in a linear problem. If a backward Euler is used, the stiffness matrix must be updated at every time step due to the dependence of this scheme on ${}^{t+\Delta t}C_n$. However, in a non-linear problem, such is the case here, there are terms in the stiffness matrix which must be updated at every time step anyway. Thus the advantages of the forward Euler and central difference schemes evaporate, and given the choice, the backward Euler is the preferred scheme to use for non-linear problems.

After discretizing the time variable, the time derivative can simply be incorporated into the already defined generic constants, i.e. since ${}^{t}C_{n}$ is known

$$\alpha_n^{\text{new}} = \alpha_n^{\text{old}} + \frac{{}^{1}C_n}{\Delta t}$$
(125)

and

$$\beta_n^{\text{new}} = \beta_n^{\text{old}} + \frac{1}{\Delta t} . \tag{126}$$

Thus in the formulation the time derivative is incorporated into the generic constants, leaving

$$0 = D_{n} \nabla^{2} C_{n} + \alpha_{n} + \beta_{n} C_{n} + \gamma_{n} C_{n}^{2} + \sum_{\substack{k=1\\k\neq n}}^{N} \rho_{nk} C_{n} C_{k} + \sum_{\substack{k=1\\k\neq n}}^{N} \eta_{k} C_{k} + \sum_{\substack{k=1\\k\neq n j\neq n}}^{N} \varepsilon_{kj} C_{k} C_{j} .$$
(127)

With the time derivative discretized, the problem can now be formulated in finite elements using the Galerkin method.

5.3 Development of the Galerkin Equations

Transformation of the original governing equation to the Galerkin equations is presented here. In later sections, the Galerkin equations will be tailored to the problem and individual terms will be identified. First the general equation is set equal to a residual term,

$$\langle R \rangle = D_{n} \nabla^{2} C_{n} + \alpha_{n} + \beta_{n} C_{n} + \gamma_{n} C_{n}^{2} + \sum_{\substack{k=1\\k\neq n}}^{N} \rho_{nk} C_{n} C_{k} + \sum_{\substack{k=1\\k\neq n}}^{N} \eta_{k} C_{k} + \sum_{\substack{k=1\\k\neq n}}^{N} \sum_{\substack{j=k\\k\neq n j\neq n}}^{N} \varepsilon_{kj} C_{k} C_{j} \quad .$$

$$(128)$$

The 'N_G' Galerkin residual equations ('N_G' will be defined later) are

$$\int_{\Omega} \langle R \rangle \phi_i d\Omega = 0 \qquad \qquad i = 1, 2, \dots, N_G \qquad (129)$$

where ϕ_i is a weighting function which will later be identified as the interpolation functions. The above provides for 'N_G' equations per bubble group (degrees of freedom). Thus the total number of equations will turn out to be 'N_G' times the maximum number of bubble groups. It will be necessary to utilize integration by parts to reduce the order of differentiation in the above equation. This manipulation is easier if the exact physical geometry is first substituted into the integral equation, i.e. 1-D spherical coordinates. Thus,

$$\mathrm{d}\Omega = 4\pi r^2 \mathrm{d}r \tag{130}$$

and

$$\nabla^2 C_n = \frac{1}{r^2} \frac{\mathrm{d}}{\mathrm{d}r} \left(r^2 \frac{\mathrm{d}C_n}{\mathrm{d}r} \right) \tag{131}$$

so that

$$\int_{\Omega} D_{n} \nabla^{2} C_{n} \phi_{i} d\Omega = 4\pi \int_{R} D_{n} \frac{d}{dr} \left(r^{2} \frac{dC_{n}}{dr} \right) \phi_{i} dr . \qquad (132)$$

The 4π is a constant that will appear in every integral and thus will have no effect on the equations. It can safely be dropped. Applying integration by parts,

$$\int_{\mathbf{R}} D_{\mathbf{n}} \frac{\mathrm{d}}{\mathrm{d}r} \left(r^2 \frac{\mathrm{d}C_{\mathbf{n}}}{\mathrm{d}r} \right) \phi_{\mathbf{i}} \mathrm{d}r = \left[\phi_{\mathbf{i}} D_{\mathbf{n}} r^2 \frac{\mathrm{d}C_{\mathbf{n}}}{\mathrm{d}r} \right]_{R_{\mathbf{i}\mathbf{n}}}^{R_{\mathbf{o}\mathbf{i}\mathbf{i}}} - \int_{\mathbf{R}} D_{\mathbf{n}} r^2 \frac{\mathrm{d}C_{\mathbf{n}}}{\mathrm{d}r} \frac{\mathrm{d}\phi_{\mathbf{i}}}{\mathrm{d}r} \mathrm{d}r \quad (133)$$

Note that the second order derivative in the original equation has been reduced to a first order derivative. The first term on the RHS of the equation is a boundary term to be evaluated at the boundaries of the element. For the problem being looked at here, the explicit boundary conditions are that the gradient in concentration is zero at the center of the grain (r = 0) and that the concentration is specified on the outer boundary. Additionally, inter-element boundary conditions require continuity of concentration and concentration gradient. For the inside element which contains a node at r = 0, the boundary term reduces to zero at r = 0 and thus requires only evaluation at the outer boundary of the element. Upon assembly of the equations and application of the specified concentration gradient is specified on a physical boundary [120]. In the problem developed here, there are no specified concentration gradients and thus integration by parts effectively yields

$$\int_{\mathbf{R}} D_{\mathbf{n}} \frac{\mathrm{d}}{\mathrm{d}r} \left(r^2 \frac{\mathrm{d}C_{\mathbf{n}}}{\mathrm{d}r} \right) \phi_{\mathbf{i}} \mathrm{d}r = -\int_{\mathbf{R}} D_{\mathbf{n}} r^2 \frac{\mathrm{d}C_{\mathbf{n}}}{\mathrm{d}r} \frac{\mathrm{d}\phi_{\mathbf{i}}}{\mathrm{d}r} \mathrm{d}r$$
(134)

with the resulting equations, which are all equal to zero, becoming

$$\int_{R} \left(-D_{n} \frac{dC_{n}}{dr} \frac{d\phi_{i}}{dr} + \phi_{i} \left[\alpha_{n} + \beta_{n} C_{n} + \gamma_{n} C_{n}^{2} + \sum_{\substack{k=1\\k=n}}^{N} \rho_{nk} C_{n} C_{k} + \sum_{\substack{k=1\\k=n}}^{N} \eta_{k} C_{k} + \sum_{\substack{k=1\\k=n}}^{N} \sum_{\substack{j=k\\k=n j=n}}^{N} \varepsilon_{kj} C_{k} C_{j} \right] \right) r^{2} dr \quad (135)$$

5.4 Incrementalization of the Galerkin Equations

It is desirable that the above equations can be solved by proven inversion matrix solving techniques. To do this the equation needs to be put into the form $[K]{C}={F}$ where [K] is the stiffness matrix, {C} is the vector containing the unknown values, and {F} is the force or load vector. This form implies that

 $[K]{C}$ should be a set of linear equations, i.e. [K] is constant with $\{C\}$ being the unknowns removed from the original equations. It is clear that putting the Galerkin equations as presented above into the form $[K]{C}={F}$ will not result in linear equations. However, several methods of dealing with this situation are available. Some of these are reviewed below. As will be seen, these are natural extensions of each other.

The simplest solution method is to assign all the constant and non-linear terms to $\{F\}$ and the coefficients to the linear terms to [K]. The non-linear terms used in $\{F\}$ would be based on values from the previous iteration, and the stiffness matrix [K] would be constant during the iterations, i.e.

$$[\mathbf{K}]^{t+\Delta t} \{\mathbf{C}\}^{i} = {}^{t+\Delta t} \{\mathbf{F}\}^{i-1}$$
(136)

where the 't' and 't+ Δ t' denote the previous time and the current time, respectively and the 'i-1' and 'i' denote the previous iteration and the current iteration, respectively. Because all the non-linearities are based entirely on previous values, the convergence rate for this method could be too slow, depending on the degree of non-linearity. In fact, very strong non-linearities might prevent convergence altogether. This method does have the advantage that **[K]** is constant and thus does not have to be updated at every iteration. Unfortunately, in the problem to be solved here the constants in **[K]** change anyway (although not at every iteration) due to explicit calculation of some of the coefficients at each time step.

Another method, which is a variation of the above, is to move all the nonlinearities from $\{F\}$ on the RHS to the LHS while factoring out one C_n , C_k , or C_j (whichever is applicable) to produce $\{C\}$ and a [K] which is partially dependent on the unknown values. Thus $\{F\}$ would remain constant and [K] would now have to be updated at every iteration with values from the previous iteration. The resulting set of equations are denoted by

$${}^{t+\Delta t}[\mathbf{K}]^{i-1} {}^{t+\Delta t}\{\mathbf{C}\}^{i} = \{\mathbf{F}\}.$$
(137)

As in the first method, the convergence rate could be too slow since the driving force towards convergence has simply been switched from $\{F\}$ to [K]. Additionally, [K] must now be recalculated at every iteration. Both of these methods require that the calculation of the unknowns at the time of interest, ${}^{t+\Delta t}C_n^{(i)}$, depend on the previously calculated values at a previous time or iteration, ${}^{t+\Delta t}C_n^{(i-1)}$ or tC_n .

A final method, and the one chosen for the current problem, is to utilize the incrementalization of C_n and cause the unknown values to be ΔC_n . An incremental dependence is achieved by defining

$${}^{t+\Delta t}C_{n}^{i} = {}^{t+\Delta t}C_{n}^{i-1} + \Delta C_{n}$$
(138)

where $\Delta C_n =$ the unknown change from the previous iteration and ${}^{t+\Delta t}C_n^{(0)} = {}^{t}C_n$. Thus if traditional Newton iteration is employed, the last calculated value from the previous time period is all that is needed to start the iteration. The resulting equations are

$${}^{t+\Delta t} [\mathbf{K}]^{i-1} {}^{t+\Delta t} \{\Delta \mathbf{C}\}^{i} = {}^{t+\Delta t} \{\mathbf{F}\}^{i-1} .$$
(139)

For an incremental, non-iterative method, the same equations are used except that there is only one iteration performed and thus the value from the previous time period, i.e. '0'th iteration, is always required. This incrementalization has the added advantage that the convergence can be easily observed since $\Delta C_n \rightarrow 0$ as convergence is achieved. The biggest advantage, however, is that the convergence is now driven by changing terms in both [K] and {F}. This accelerates convergence as well as provides additional stability for strong non-linearities. In fact, this third implementation can be considered to be a combination of the first two methods, incorporating the best of both (at least in the problem to be solved here). Again [K] must be recalculated at each iteration, but the gains in stability and convergence rate are considered to be worth the increase in calculations.

It should be obvious from the above discussions that the values at time 't+ Δ t' are always dependent on previously iterated values at time 't+ Δ t' except for the very first iteration, where the values from time 't' should be used. With this understood, the writing of the basic equations can be simplified by dropping the time superscript. Additionally, it is apparent that if the unknowns are incrementalized as defined above, then only the 'i-1' designator appears. Thus the iteration superscript can also be dropped with the understanding that no superscript refers to the last known value, i.e. the 'i-1' th iteration. Except for a few obvious places, both these superscripts will be dropped for further illustrations, i.e. $C_n = {}^{t+\Delta t}C_n^{(i-1)}$.

Using this modified notation, the incrementalized concentration terms can now be substituted into the Galerkin equations. After some rearrangement the Galerkin equations become

$$\int_{R} \left\{ \left[-D_{n} \frac{d}{dr} \frac{d\phi_{i}}{dr} + \phi_{i} \left(\beta_{n} + 2\gamma_{n}C_{n} + \sum_{\substack{k=1 \ k \neq n}}^{N} \rho_{nk}C_{k} \right) \right] \Delta C_{n} + \phi_{i} \left[\sum_{\substack{k=1 \ k \neq n}}^{N} \rho_{nk}C_{n} + \sum_{\substack{k=1 \ k \neq n}}^{N} \eta_{k} + \sum_{\substack{k=1 \ j = k}}^{N} \varepsilon_{kj}C_{j} \right] \Delta C_{k} + \phi_{i} \left[\sum_{\substack{k=1 \ j = k}}^{N} \sum_{\substack{k=1 \ k \neq n}}^{N} \varepsilon_{kj}C_{k} \right] \Delta C_{j} \right] r^{2} dr + \int_{R} \left\{ \phi_{i} \left[\gamma_{n} (\Delta C_{n})^{2} + \sum_{\substack{k=1 \ k \neq n}}^{N} \rho_{nk} \Delta C_{n} \Delta C_{k} + \sum_{\substack{k=1 \ k \neq n}}^{N} \sum_{\substack{j=k \ k \neq n \ j \neq n}}^{N} \varepsilon_{kj} \Delta C_{j} \right] \right\} r^{2} dr = \int_{R} \left\{ D_{n} \frac{dC_{n}}{dr} \frac{d\phi_{i}}{dr} - \phi_{i} \left[\alpha_{n} + \beta_{n}C_{n} + \gamma_{n}C_{n}^{2} + \sum_{\substack{k=1 \ k \neq n}}^{N} \rho_{nk}C_{n}C_{k} + \sum_{\substack{k=1 \ k \neq n}}^{N} \eta_{n}C_{k} + \sum_{\substack{k=1 \ k \neq n}}^{N} \varepsilon_{kj}C_{k}C_{j} \right] \right\} r^{2} dr$$

$$(140)$$

Note that the unknown quadratic (non-linear) terms have been separated out on the LHS. Except for these terms the above equation would easily fit into the form $[K]{\Delta C}={F}$. Inspection of the above equations will reveal that the LHS controls the rate of convergence while the RHS controls the accuracy of the result. Additionally, as convergence is approached the quadratic terms become negligible very quickly. Thus it can safely be assumed that dropping the quadratic terms will only slow the convergence rate and will have no effect on the resulting solution. The equation can further be simplified by noting that the 'k' and 'j' indices in the double summation terms of the LHS can be interchanged without affecting the equation. Making these changes yields the equation

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$$\int_{R} \left\{ \left[-D_{n} \frac{d}{dr} \frac{d\phi_{i}}{dr} + \phi_{i} \left(\beta_{n} + 2\gamma_{n}C_{n} + \sum_{\substack{k=1\\k\neq n}}^{N} \rho_{nk}C_{k} \right) \right] \Delta C_{n} + \phi_{i} \left[\sum_{\substack{k=1\\k\neq n}}^{N} \rho_{nk}C_{n} + \sum_{\substack{k=1\\k\neq n}}^{N} \eta_{k} + \sum_{\substack{k=1\\k\neq n}}^{N} \sum_{\substack{j=k\\k\neq n \neq n}}^{N} \varepsilon_{kj}C_{j} + \sum_{\substack{j=1\\j\neq n}}^{N} \sum_{\substack{k=1\\j\neq n k\neq n}}^{N} \varepsilon_{jk}C_{j} \right] \Delta C_{k} \right\} r^{2} dr = \int_{R} \left\{ D_{n} \frac{dC_{n}}{dr} \frac{d\phi_{i}}{dr} - \phi_{i} \left[\alpha_{n} + \beta_{n}C_{n} + \gamma_{n}C_{n}^{2} + \sum_{\substack{k=1\\k\neq n}}^{N} \rho_{nk}C_{n}C_{k} + \sum_{\substack{k=1\\k\neq n}}^{N} \eta_{n}C_{k} + \sum_{\substack{k=1\\k\neq n \neq n}}^{N} \varepsilon_{kj}C_{k}C_{j} \right] \right\} r^{2} dr$$

$$(141)$$

The '[K]', ' $\{\Delta C\}$ ', and ' $\{F\}$ ' terms are now readily identifiable, i.e.

$$[\mathbf{K}]\{\Delta \mathbf{C}\} = \{\mathbf{F}\} \Rightarrow \begin{bmatrix} K_{11} \cdots K_{1N} \\ \vdots & \ddots & \vdots \\ K_{N1} \cdots K_{NN} \end{bmatrix} \begin{bmatrix} \Delta C_1 \\ \vdots \\ \Delta C_N \end{bmatrix} = \begin{bmatrix} F_1 \\ \vdots \\ F_N \end{bmatrix}$$
(142)

where

$$K_{nn} = \int_{R} \left\{ -D_{n} \frac{\mathrm{d}}{\mathrm{d}r} \frac{\mathrm{d}\phi_{i}}{\mathrm{d}r} + \phi_{i} \left(\beta_{n} + 2\gamma_{n}C_{n} + \sum_{\substack{k=1\\k=n}}^{N} \rho_{nk}C_{k} \right) \right\} r^{2} \mathrm{d}r$$
(143)

and

$$K_{nk} = \int_{R} \left\{ \phi_i \left(\rho_{nk} C_n + \eta_k + \sum_{\substack{j=k \ j\neq n}}^{N} \varepsilon_{kj} C_j + \sum_{\substack{j=n \ j\neq k}}^{N} \varepsilon_{jk} C_j \right) \right\} r^2 dr .$$
 (144)

The rest of the matrix terms not identified are filled out in a similar manner. The terms above are repeated for every weighting function, ϕ_i . The final stiffness matrix [K] will consist of 'N_G× N' rows. Discretization of the problem allows one to determine the weighting functions and causes the stiffness matrix to become square with 'N_G× N' columns.

5.5 Discretization of the Galerkin Equations

So that the Galerkin equations can be formulated in a form useful for solving real problems, the specific problem must be discretized. Discretization is a necessary prerequisite to determining weighting functions and methods to calculate values such as ' C_n ' and ' ΔC_n '. First, the problem is attacked in a general form by assuming physical elements with 'NG' nodes per element. Normally, the first and last nodes (in a 1-D problem) determine the physical boundaries of the element. Within each element, the values at the nodes will be known or calculated. Between nodes, values will have to be interpolated using weighting functions (akainterpolation functions), i.e. the ϕ_i 's which were used above. The requirements for interpolating functions are that ϕ_i equals one at the i'th node and equals zero at all other nodes. There are no requirements governing the behavior of the weighting functions between nodes, and continuity across element boundaries is automatically taken care off by the formation of the Galerkin equations themselves. So that the nodal values can be differentiated from the interpolated values, a subscript modification will be introduced which refers to the local (elemental) node referred to, i.e. $C_{n:i}$ will refer to a nodal concentration where 'n' indicates the bubble group, as before, and 'i' indicates the node number. The interpolated values are related to the nodal values by

$$C_{n} = \sum_{i=1}^{N_{G}} \phi_{i} C_{n:i}$$
 (145)

and

$$\Delta C_{n} = \sum_{i=1}^{N_{G}} \phi_{i} \Delta C_{n:i}$$
(146)

and

$$r_{\rm n} = \sum_{i=1}^{N_{\rm G}} \phi_i r_{\rm n:i}$$
 (147)

These values are substituted into the previously developed Galerkin equations. Because of the interpolation of ΔC_n , every degree of freedom in an element generates 'N_G' unknowns to be determined in the discretized Galerkin equations. The result is a [K] matrix that is square of dimensions 'N_G× N' as well as { ΔC } and {F} vectors of length 'N_G× N'. Note that because of the first order gradient term, the weighting functions must be continuous and once differentiable. Had iteration upon the original governing equation been formulated instead of a Galerkin formulation, the weighting functions would have had to be twice differentiable.

Once the elemental matrices and vectors have been formed, they are assembled into a global matrix and vectors which are solved using common solving routines such as Gaussian elimination or direct substitution-iteration. For the problem here, Gaussian elimination was chosen as the solving method.

5.6 Determination of Interpolating Functions

Interpolation functions are used to interpolate between nodes in a problem. When elements are defined within a problem, the interpolation functions need only interpolate within an element. Values between elements are accounted during assembly into the global matrix through the boundary nodes connecting elements. In theory, one would want interpolation functions which exactly fit the profile of the unknown being modeled. In practice, this cannot be done since one must know the solution to know the profile. However, a general idea of the solution profile is usually known, and appropriate interpolation functions can be chosen. The only criteria for the i'th interpolation function is that it is has a specified value (usually one) at the i'th node and is zero at all other elemental nodes. Other constraints can

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be imposed, such as requiring orthogonal functions, but are not required. Since most problem solutions are continuous and smoothly varying, the most often used interpolation functions are based on simple polynomials. More complex functions have been used when the problem solution warrants it. However, it becomes increasingly more difficult to meet the nodal criteria mentioned above with more complex functions. Another method used to keep the interpolation functions simple is to keep the number of elemental nodes low. Thus, typically, investigators use only two or three nodes per element per dimension. The polynomial interpolation functions for both these sets are presented in fig. 1. The specific characteristics of the problem solution here are best modeled using three nodes per element in the outermost element. For this reason, program REDSTONE was programmed to use three nodes per element for all elements.

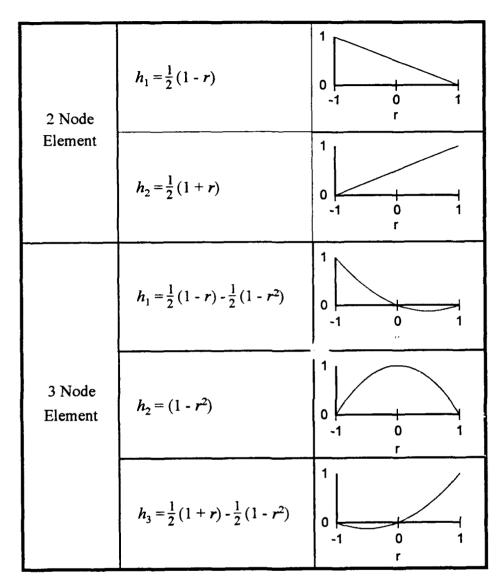


Fig. 1 Interpolation Functions for Two and Three Nodes

CHAPTER VI BUBBLE RADII FORMULATION

6.1 Generalities

The finite element formulation for the bubble radii equations involves more detail than for the concentration equations due to higher non-linearities and the fact that some of the fuel properties depend on bubble radius. Specific dependencies were discussed in Chapter IV and included the resolution parameter, source term, diffusion coefficient, pressure, and bubble velocity. It was also mentioned in that chapter that all but one of the dependencies, pressure, could be left out of the actual finite element implementation in program REDSTONE. The bulk of the formulation method for bubble radii remains the same as for concentrations. In fact, the main difference is the apportionment of terms in the governing equation to the correct finite element terms. In Chapter IV, use was made of a generic governing equation including constant, linear, and quadratic terms. This equation was then incrementalized before being put into Galerkin residual equations. Because of the higher non-linearities, the generic equation in Chapter IV is not sufficient for describing the bubble radii equations.

In this chapter, rather than approach the formulation using a general equation, each individual process will be looked at separately. These processes are the same ones previously discussed. Each process will be incrementalized, leaving only constant and linear terms to deal with. Two incrementalizations are presented, one with full radius dependence incorporated and one with the radius dependent properties and concentration equation terms held constant.. Except for pressure, this latter form was the one used in program REDSTONE. Once incrementalized, the methods used in Chapter IV to build the Galerkin equations, the stiffness matrix, and force vector are the same needed for bubble radii equations. The incrementalized bubble radii terms need only be put into residual Galerkin equations, with linear terms being put into the stiffness matrix and constant terms being put into the force vector. Only the diffusion gradient term needs to be specifically looked at in terms of the formulation of the residual Galerkin equations. This occurs because the bubble radius, which now modifies the diffusion term, is also a function of position in the grain.

Extensive use is made of expansion in Taylor series which have been truncated after first order terms, i.e.

$$f(x + \Delta x) \approx f(x) + \Delta x \frac{f'(x)}{1!} . \qquad (148)$$

The purpose of using Taylor series is to isolate unknowns in linear, first order terms, yet keep the non-linear behavior. In practice, the formulation method used here could have been used for the concentration formulation with exactly the same results.

The bubble radii adjustment equation was presented in Chapter III. It is reprinted here for reference,

$$C_{n} \frac{\partial R_{n}}{\partial t} = \left(D_{n} \nabla^{2} C_{n} \right) \frac{R_{n}}{3} - K_{n} C_{n} \frac{R_{n}}{3} + K_{n-1} C_{n-1} \frac{R_{n-1}^{3}}{3R_{n}^{2}} - b_{n} C_{n} N_{n} \frac{R_{n}}{3} + b_{n+1} C_{n+1} N_{n+1} \frac{R_{n+1}^{3}}{3R_{n}^{2}} - 32\pi R_{n} D_{n} C_{n}^{2} \frac{R_{n}}{3} - \sum_{\substack{k=1\\k\neq n}}^{N} \left[4\pi (R_{n} + R_{k}) (D_{n} + D_{k}) - \pi (R_{n} + R_{k})^{2} |V_{n} - V_{k}| \right] C_{n} C_{k} \frac{R_{n}}{3} + \sum_{\substack{k=1\\k\neq n}}^{N/2} \left[4\pi (R_{n-k} + R_{k}) (D_{n-k} + D_{k}) - \pi (R_{n-k} + R_{k})^{2} |V_{n-k} - V_{k}| \right] C_{n-k} C_{k} \left(\frac{R_{k}^{3}}{3R_{n}^{2}} + \frac{R_{n-k}^{3}}{3R_{n}^{2}} \right) - \frac{\Omega}{R_{n}} D_{i} C_{i} C_{n} + \frac{\Omega}{R_{n}} D_{v} C_{n} \left\{ C_{v} - C_{v}^{e} \exp \left[-\frac{\Omega}{kT} \left(P_{n} - \frac{2\gamma}{R_{n}} - P_{o} \right) \right] \right\} \right\}.$$
(149)

As programmed in REDSTONE, the radii which are changed during radii calculations are the ones at the very ends of the terms and the vacancy/interstitial interaction terms. Other radii remain constant during calculations.

6.2 Incrementalization of Individual Processes

6.2.1 Resolution Loss

Resolution loss is described by

Resolution Loss =
$$\frac{1}{3}b_nC_nR_nN_n$$
 (150)

where the resolution parameter, b_n , for bubbles with $R_n > 15$ Å is given by

$$b_{\rm n} = b\eta_{\rm n} = b\left\{1 - \left(X^3 - \frac{3X^2Y}{R_{\rm n}} + \frac{3XY^2}{R_{\rm n}^2} - \frac{Y^3}{R_{\rm n}^3}\right)\right\}$$
(151)

where

$$X = 1 - \frac{Y}{B} \left(\frac{kT}{2\gamma}\right) \tag{152}$$

$$Y = 15 \times 10^8 .$$
 (153)

Combining this radius dependency with the loss term, incrementalizing R_n , and applying Taylor series gives

$$\text{Loss} = \frac{1}{3}b_{n}C_{n}N_{n}R_{n} + \frac{1}{3}bC_{n}N_{n}\left\{1 - \left(X^{3} - \frac{3X^{2}Y}{R_{n}} + \frac{3XY^{2}}{R_{n}^{2}} - \frac{2Y^{3}}{R_{n}^{3}}\right)\right\}\Delta R_{n} \quad (154)$$

Note that the use of incrementalization and Taylor series have converted the resolution loss term into the last known loss term plus the correction term for the new time step. The form above would only be applied for bubbles with $R_n > 15$ Å For smaller bubbles, b_n has no radius dependence. Such a discrepancy could introduce oscillations into the radius calculations if a bubble has a radius approximately equal to 15 Å The corresponding incrementalized loss term for small bubbles is

$$Loss = \frac{1}{3}b_n C_n N_n \left(R_n + \Delta R_n\right)$$
(155)

which is considerably simplified. This is also the form used if one leaves out the radius dependence in the resolution parameter altogether and is the form used in program REDSTONE.

6.2.2 Resolution Gain

Resolution gain is somewhat more difficult to handle than loss because of the mixture of dependencies on two different bubble radii. However, the basic approach for incrementalization is the same. The gain term is given by

Resolution Gain =
$$\frac{1}{3}b_k C_k N_k \frac{R_k^3}{R_n^2}$$
 (156)

where the 'n+1' subscript has been replaced by a 'k' for simplification. Using the same methods as for resolution loss, one arrives at (for $R_k>15$ Å),

$$Gain = \frac{1}{3}b_{k}C_{k}N_{k}\frac{R_{k}^{3}}{R_{n}^{2}} + bC_{k}N_{k}\left\{\left(1 - X^{3}\right)\frac{R_{k}^{2}}{R_{n}^{2}} + 2X^{2}Y\frac{R_{k}}{R_{n}^{2}} - XY^{2}\frac{1}{R_{n}^{2}}\right\}\Delta R_{k}$$
$$+ bC_{k}N_{k}\left\{\frac{2}{3}\left(X^{3} - 1\right)\frac{R_{k}^{3}}{R_{n}^{3}} - 2X^{2}Y\frac{R_{k}^{2}}{R_{n}^{3}} + 2XY^{2}\frac{R_{k}}{R_{n}^{3}} - \frac{2}{3}Y^{3}\frac{R_{k}^{3}}{R_{n}^{5}}\right\}\Delta R_{n} \quad (157)$$

For smaller bubbles and retaining the radius dependence only partially,

$$Gain = \frac{1}{3}b_k C_k N_k \left[\frac{R_k^3}{R_n^2} + \frac{3R_k^2}{R_n^2} \Delta R_k - \frac{2R_k^3}{R_n^3} \Delta R_n \right]$$
(158)

which is the form used in REDSTONE.

6.2.3 Self-Coalescence Loss

Loss through coalescence of bubbles with same size bubbles is given by

Self - coalescence Loss =
$$\frac{32}{3}\pi R_n^2 D_n C_n^2$$
 (159)

The diffusion coefficient radius dependence can be expressed by

$$D_{\rm n} = \frac{D_{\rm o}}{R_{\rm n}^2} \tag{160}$$

then the loss term becomes

Self - coalescence Loss =
$$\frac{32}{3}\pi D_o C_n^2$$
 (161)

where

$$D_{\rm o} = \frac{6\lambda\Omega D_{\rm s}}{\pi\lambda_{\rm s}^2} \sin^2\left(\frac{\lambda_{\rm s}}{2R_{\rm n}}\right) \qquad \qquad R_{\rm n} > \frac{\lambda_{\rm s}}{\pi} \qquad (162)$$

$$D_{\rm o} = \frac{6\lambda\Omega D_{\rm s}}{\pi\lambda_{\rm s}^2} \qquad \qquad R_{\rm n} \le \frac{\lambda_{\rm s}}{\pi} \ . \tag{163}$$

The sine term was not incorporated explicitly into the formulation even though it depends on R_n . This was done because of problems with using Taylor series for a periodic function and because of the conditional application of the term. However, the sine term is still accounted for in the last known solution incorporated into the

force vector. Note also that without the sine term, the resulting incrementalized loss term has no dependence on bubble radius.

If, instead, the diffusion coefficient dependence is left out and the original term from the concentration equation is held constant, then the incremental solution becomes

Self - coalescence Loss =
$$\frac{32}{3}\pi R_n D_n C_n^2 [R_n + \Delta R_n]$$
 (164)

where only the last term in brackets changes during calculation. This is the form used in REDSTONE.

6.2.4 Coalescence Loss

Loss through coalescence of bubbles with different size bubbles is given by Coalescence Loss = $\frac{4}{3}\pi R_n (R_n + R_k) (D_n + D_k) C_n C_k$. (165) Applying methods as before with full radius dependence included yields,

Coalescence Loss =
$$\frac{4}{3}\pi R_n (R_n + R_k) (D_n + D_k) C_n C_k$$

+ $\frac{4}{3}\pi C_n C_k [D_k (2R_n + R_k) - D_n R_k] \Delta R_n$
+ $\frac{4}{3}\pi C_n C_k R_n \left[D_n - D_k \left(1 + 2\frac{R_n}{R_k} \right) \right] \Delta R_k$ (166)

and with radius dependence partially left out,

Coalescence Loss =
$$\frac{4}{3}\pi(R_n + R_k)(D_n + D_k)C_nC_k[R_n + \Delta R_n]$$
 (167)

where only the last term in brackets changes during calculation.

6.2.5 Coalescence Gain

Gain through coalescence of two smaller bubbles is given by

Coalescence Gain =
$$\frac{4}{3}\pi (R_j + R_k) (D_j + D_k) C_j C_k \frac{1}{R_n^2} (R_j^3 + R_k^3)$$
 (168)

where the 'n-k' subscript has been replaced with 'j' for convenience. Applying methods as before with full radius dependence included yields,

Coalescence Gain =
$$\frac{4}{3}\pi (R_{j} + R_{k})(D_{j} + D_{k})C_{j}C_{k}\frac{1}{R_{n}^{2}}(R_{j}^{3} + R_{k}^{3})$$

 $-\frac{8}{3}\pi C_{j}C_{k}\frac{1}{R_{n}^{3}}(R_{j}^{3} + R_{k}^{3})(R_{j} + R_{k})(D_{j} + D_{k})\Delta R_{n}$
 $+\frac{4}{3}\pi C_{j}C_{k}\frac{1}{R_{n}^{2}}\left[\frac{D_{k}}{R_{k}}(2R_{k}^{4} - R_{j}^{3}R_{k} + R_{j}R_{k}^{3} - 2R_{j}^{4})$
 $+D_{j}(4R_{k}^{3} + R_{j}^{3} + 3R_{k}^{2}R_{j})\right]\Delta R_{k}$
 $+\frac{4}{3}\pi C_{j}C_{k}\frac{1}{R_{n}^{2}}\left[\frac{D_{j}}{R_{j}}(2R_{j}^{4} - R_{k}^{3}R_{j} + R_{k}R_{j}^{3} - 2R_{k}^{4})$
 $+D_{k}(4R_{j}^{3} + R_{k}^{3} + 3R_{j}^{2}R_{k})\right]\Delta R_{j}$ (169)

and with radius dependence partially left out,

Coalescence Gain =
$$\frac{4}{3}\pi (R_j + R_k) (D_j + D_k) C_j C_k$$

 $\times \left[\frac{R_k^3 + R_j^3}{R_n^2} - 2 \frac{R_k^3 + R_j^3}{R_n^3} \Delta R_n + \frac{3R_k^2}{R_n^2} \Delta R_k + \frac{3R_j^2}{R_n^2} \Delta R_j \right]$ (170)

where only the last term in brackets changes during calculation.

6.2.6 Biased Coalescence Loss

Loss through biased coalescence with another bubble is given by

Biased Coalescence Loss =
$$\frac{1}{3}\pi R_n (R_n + R_k)^2 |V_n - V_k| C_n C_k$$
 (171)

Bubble velocity depends on radius with the dependency,

$$V_{\rm n} = V_{\rm o} R_{\rm n}^3$$
 (172)

Unfortunately, incrementalization of the velocity requires splitting the terms into constant and correction pieces which are then placed at different locations in the stiffness matrix and force vector. Keeping the absolute value incorporated becomes extremely difficult. Because of this, the velocity terms are kept constant in both the radius dependent and radius independent formulations. The absolute value then resides in the force vector during calculations. However, incorporating the other radius dependent terms into the formulation gives,

Biased Coalescence Loss =
$$\frac{1}{3}\pi R_n (R_n + R_k)^2 |V_n - V_k| C_n C_k$$

+ $\frac{1}{3}\pi (3R_n^2 + 4R_n R_k + R_k^2) |V_n - V_k| C_n C_k \Delta R_n$
+ $\frac{1}{3}\pi (2R_n^2 + 2R_n R_k) |V_n - V_k| C_n C_k \Delta R_k$. (173)

Holding radius dependent terms from the concentration equation constant yields the form used in REDSTONE,

Biased Coalescence Loss = $\frac{1}{3}\pi (R_n + R_k)^2 |V_n - V_k| C_n C_k [R_n + \Delta R_n]$ (174) where only the last term in brackets changes during calculation.

6.2.7 Biased Coalescence Gain

Gain through biased coalescence of two smaller bubbles is given by

Biased Coalescence Gain =
$$\frac{1}{3}\pi (R_j + R_k)^2 |V_j - V_k| C_j C_k \frac{1}{R_n^2} (R_j^3 + R_k^3)$$
 (175)

where the 'n-k' subscript has again been replaced with 'j' for convenience. The arguments for keeping the velocity terms constant in biased coalescence loss also apply here. Applying methods as before to the rest of the radius dependence in the above yields

Biased Coalescence Gain =
$$\frac{1}{3}\pi (R_{j} + R_{k})^{2} |V_{j} - V_{k}| C_{j}C_{k}\frac{1}{R_{n}^{2}}(R_{j}^{3} + R_{k}^{3})$$

 $-\frac{2}{3}\pi (R_{j} + R_{k})^{2} |V_{j} - V_{k}| C_{j}C_{k}\frac{1}{R_{n}^{3}}(R_{j}^{3} + R_{k}^{3})\Delta R_{n}$
 $+\frac{\pi}{3R_{n}^{2}}(5R_{k}^{4} + 8R_{k}^{3}R_{j} + 3R_{k}^{2}R_{j}^{2} + 2R_{k}R_{j}^{3} + 2R_{j}^{4})|V_{j} - V_{k}|C_{j}C_{k}\Delta R_{k}$
 $+\frac{\pi}{3R_{n}^{2}}(5R_{j}^{4} + 8R_{j}^{3}R_{k} + 3R_{j}^{2}R_{k}^{2} + 2R_{j}R_{k}^{3} + 2R_{k}^{4})|V_{j} - V_{k}|C_{j}C_{k}\Delta R_{j}$
(176)

and holding the radius dependent terms carried over from the concentration equation constant yields

Biased Coalescence Gain =
$$\frac{1}{3}\pi \left(R_{j} + R_{k}\right)^{2} \left|V_{j} - V_{k}\right| C_{j}C_{k}$$

 $\times \left[\frac{R_{j}^{3} + R_{k}^{3}}{R_{n}^{2}} - 2\frac{R_{j}^{3} + R_{k}^{3}}{R_{n}^{3}}\Delta R_{n} + \frac{3R_{k}^{2}}{R_{n}^{2}}\Delta R_{k} + \frac{3R_{j}^{2}}{R_{n}^{2}}\Delta R_{j}\right]$ (177)

where only the last term in brackets changes during calculation.

6.2.8 Source Loss

Loss through deposition of another fission gas atom in the bubble is given by Source Loss = $\frac{1}{3}K_nC_nR_n$ (178)

where the radius dependence of the source term is

$$K_{\rm n} = K_{\rm o} R_{\rm n}^2$$
 (179)

Inclusion of this into the incremental formulation gives

Source Loss =
$$\frac{1}{3}K_nC_nR_n + K_nC_n\Delta R_n$$
 (180)

or if the source radius dependence is left out,

Source Loss =
$$\frac{1}{3}K_nC_n[R_n + \Delta R_n]$$
 (181)

where only the last term in brackets changes during calculation.

6.2.9 Source Gain

Gain through deposition of another fission gas atom in the next smaller bubble is given by

Source Gain =
$$\frac{1}{3}K_k C_k \frac{R_k^3}{R_n^2}$$
 (182)

where the 'n-1' subscript has been replaced by 'k' for convenience. The radius dependence of the source term is as stated in the last section. The full radius dependent formulation is

Source Gain =
$$\frac{1}{3}K_kC_k\frac{R_k^3}{R_n^2} - \frac{2}{3}K_kC_k\frac{R_k^3}{R_n^3}\Delta R_n + \frac{5}{3}K_kC_k\frac{R_k^2}{R_n^2}\Delta R_k$$
 (183)

with the partially radius independent formulation being

Source Gain =
$$\frac{1}{3}K_k C_k \left[\frac{R_k^3}{R_n^2} - 2\frac{R_k^3}{R_n^3} \Delta R_n + 3\frac{R_k^2}{R_n^2} \Delta R_k \right]$$
 (184)

where only the term in brackets changes during calculation.

6.2.10 Interstitial Capture

A decrease in radius due to capture of interstitials is given by

Interstitial Capture Decrease =
$$\Omega D_i C_i C_n \frac{1}{R_n}$$
. (185)

The incremental formulation of this term is

Interstitial Capture Decrease =
$$\frac{\Omega D_i C_i C_n}{R_n} \left[1 - \frac{\Delta R_n}{R_n} \right]$$
. (186)

This is the form used in REDSTONE. There is not a partially radius dependent form as this term did not appear in the original concentration equation.

6.2.11 Vacancy Capture

An increase in radius due to capture of vacancies is given by

Vacancy Capture Increase =
$$\Omega D_v C_v C_n \frac{1}{R_n}$$
 (187)

The incremental formulation of this term is similar to that for interstitial capture, i.e.

Vacancy Capture Increase =
$$\frac{\Omega D_v C_v C_n}{R_n} \left[1 - \frac{\Delta R_n}{R_n} \right]$$
 (188)

Again, there is not a partially radius dependent form as this term did not appear in the original concentration equation.

6.2.12 Vacancy Emission

A decrease in radius due to emission of vacancies is given by

Vacancy Emission Decrease =
$$\Omega D_v C_n \frac{1}{R_n} C_v^c \exp\left\{\frac{\Omega}{kT}\left(P_o + \frac{2\gamma}{R_n} - P_n\right)\right\}$$
. (189)

The incremental formulation of this term is quite different from the previous two terms because of the exponential which is dependent on both bubble radius and bubble pressure. The bubble pressure has a radius dependence given by

$$P_{\rm n} = \frac{X}{R_{\rm n}^3 - Y}$$
(190)

where

$$X = \frac{4N_{\rm n}kT}{3\pi} \tag{191}$$

and

$$Y = \frac{4N_{\rm n}B}{3\pi} \ . \tag{192}$$

After incrementalization and then Taylor series expansion, one obtains

Vacancy Emission Decrease =
$$\frac{\Omega D_{\rm v} C_{\rm n}}{R_{\rm n}} C_{\rm v}^{\rm e} \exp\left\{\frac{\Omega}{kT} \left(P_{\rm o} + \frac{2\gamma}{R_{\rm n}} - P_{\rm n}\right)\right\}$$
$$\times \left[1 - \left(\frac{1}{R_{\rm n}} + \frac{2\Omega\gamma}{kTR_{\rm n}^2} - \frac{3\Omega XR_{\rm n}^2}{kT\left(R_{\rm n}^3 - Y\right)^2}\right)\right] \Delta R_{\rm n}$$
(193)

which is the form used in REDSTONE.

6.3 Modification of the Diffusion Term

The diffusion term for bubble radius adjustment is given by

Diffusion Loss =
$$\frac{R_n D_n}{3r^2} \frac{d}{dr} \left(r^2 \frac{dC_n}{dr} \right)$$
 (194)

Before incrementalizing this term, it will be useful to first incorporate it into a Galerkin formulation. Holding the diffusion coefficient constant, the Galerkin residual equations for this term are given by

Diffusion Loss =
$$\frac{4\pi}{3} D_n \int_{\mathbb{R}} R_n \frac{\mathrm{d}}{\mathrm{d}r} \left(r^2 \frac{\mathrm{d}C_n}{\mathrm{d}r} \right) \phi_n \mathrm{d}r$$
 (195)

As before, the 4π can be dropped. The above term is different from the formulation in the previous chapter by the presence of R_n . Applying integration by parts gives

Diffusion Loss =
$$\left[\frac{1}{3}D_nR_n\phi_ir^2\frac{dC_n}{dr}\right]_{R_{out}}^{R_{in}} - \frac{D_n}{3}\int_{R}r^2\frac{dC_n}{dr}\frac{d(R_n\phi_i)}{dr}dr$$
 (196)

The first term in brackets is evaluated at the element boundaries. Like the concentration gradient, the concentration gradient times the bubble radius must be continuous and smoothly varying across the boundary. There can be no jumps in bubble size at a boundary. As a result, this term cancels out everywhere except on physical boundaries where the gradient is specified. Since the gradient is never

specified in the problem here, this term can be eliminated entirely. The second term involves the gradient of a product. Expanding this out gives for the diffusion loss term

Diffusion Loss =
$$-\frac{D_n}{3} \int_{R} R_n r^2 \frac{dC_n}{dr} \frac{d\phi_i}{dr} dr - \frac{D_n}{3} \int_{R} \phi_i r^2 \frac{dC_n}{dr} \frac{dR_n}{dr} dr$$
 (197)

The result of the R_n has been to add an extra term to be included into the formulation. These terms can now be incrementalized and incorporated using the techniques presented in Chapter V.

CHAPTER VII PROGRAM REDSTONE ORGANIZATION

7.1 Overview

Program REDSTONE is organized as shown in fig. 2. The flow of program control involves three separate and one overall iteration cycles. The three separate iteration cycles involve calculations of bubble concentrations, bulk vacancy and interstitial concentrations, and bubble radii. The overall iteration cycle is completed when no additional iterations are required in the bubble concentration or bubble radii steps during any one overall iteration. Time stepping is done using time steps and time periods. Periods define a set of operational conditions for a length of time. Time steps are user specified equal interval divisions within the time period to facilitate calculational and/or output requirements. Transient analysis is accomplished by stepping between the initial and final period conditions based on the number of time steps in a period. Within REDSTONE is built-in the capability to 'restart' from a previously saved analysis. This allows one to set up a specific initial profile for subsequent exploratory analyses. A complete program listing can be found in Appendix 1.

Input to the program is accomplished through an input file with data in free format and identified by position on a line and a unique line number. Comment lines are allowed and identified by an '*' or 'C' at the beginning of the line. Optionally, a restart input file is also used to input starting conditions. Output of the program consists of readable output at user-specified intervals, and a character based file for input to separate plotting/data analysis schemes. Additionally, after each

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period, specific parameters are saved in machine readable form for use in subsequent restart analyses using program REDSTONE.

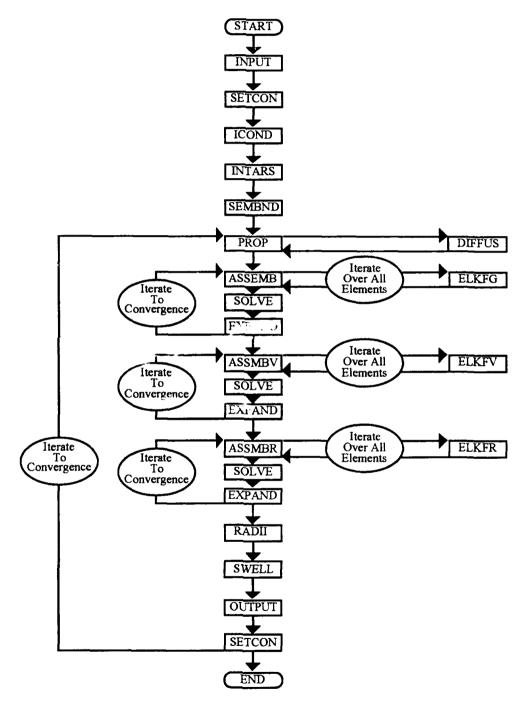


Fig. 2 REDSTONE Program Flow.

7.2 Subroutines

Program REDSTONE is made up of 27 subroutines and functions. Each of these are listed below with a short description of their functions.

7.2.1 ASSEMB

Assembles the global matrix and force vector to be solved for bubble concentrations. The global stiffness matrix is produced in a banded, non-symmetrical form with specified concentration rows and columns (those on the grain surface) eliminated but still accounted for. The bandwidth of the global matrix is twice the number of bubble sizes times the number of nodes in an element.

7.2.2 ASSMBR

Assembles the global matrix and force vector to be solved for bubble radii. Again, the global stiffness matrix is produced in a banded, non-symmetrical form with specified radii rows and columns (those on the grain surface and gas atoms) eliminated but still accounted for. The bandwidth of the global matrix is twice the number of bubble sizes minus one times the number of nodes in an element. The number of bubble sizes is reduced by one because the radius of single gas atoms is not variable.

7.2.3 ASSMBV

Assembles the global matrix and force vector to be solved for bulk vacancy and interstitial concentrations. The global stiffness matrix is produced in a banded, non-symmetrical form with specified radii rows and columns (those on the grain surface) eliminated but still accounted for. The bandwidth of the global matrix is twice the number of nodes per element times two.

7.2.4 DIFFUS

Calculates diffusion coefficients for single gas atoms, bubbles, vacancies, and interstitials.

7.2.5 ELKFG

Assembles the elemental stiffness matrices and force vectors of bubble concentrations for subsequent integration into the global matrix and vector by ASSEMB. The governing equations are incorporated in this subroutine.

7.2.6 ELKFR

Assembles the elemental stiffness matrices and force vectors of bubble radii for subsequent integration into the global matrix and vector by ASSMBR. The governing equations are incorporated in this subroutine.

7.2.7 ELKFV

Assembles the elemental stiffness matrices and force vectors of vacancy and interstitial concentrations for subsequent integration into the global matrix and vector by ASSMBV. The governing equations are incorporated in this subroutine.

7.2.8 EQURAD

Calculates the equilibrium radius of a given bubble size with temperature and gas atom content as inputs.

7.2.9 EXPAND

Adds the iterative solutions into the respective change vectors describing the change over a time period.

7.2.10 ERROR

Prints the user specified error in the output.

7.2.11 FRTPAG

Prints the REDSTONE banner in the output.

7.2.12 ICOND

Calculates arrays describing connectivity between bubble groups at elemental nodes with bubble groups at global nodes. Used by ASSEMB and ASSMBR.

7.2.13 INPUT

Assigns data read in from the input file to respective variables.

7.2.14 INTARS

Calculates arrays describing correlations of all bubble groups to those which are not specified. Used by ASSEMB and ASSMBR.

7.2.15 MTXADD

Performs matrix addition on arrays.

7.2.16 OUTPUT

Prints results of calculations in character format.

7.2.17 PROP

Calculates fuel, fission gas, and bubble properties at the beginning of each new time step. Also calculates initial bubble radii.

7.2.18 RADII

Calculates new bubble radii based on calculation results.

7.2.19 RDCARD

Reads a line of input data from the input file and parses it.

7.2.20 RESINP

Reads the restart input file when restart is specified.

7.2.21 **RESSAV**

Saves data in machine readable form to a restart file after each time period. Allows subsequent runs to start at a specific point in time and with an already calculated concentration and radii profile.

7.2.22 RNORM

Calculates the norm of a vector.

7.2.23 SEMBND

Calculates the half-bandwidth of the bubble concentration stiffness matrix. Similar values for vacancy/interstitial concentrations and bubble radii are calculated in assembly routines.

7.2.24 SETCON

Adds the time period changes into the known values and sets changes to zero.

7.2.25 SHAPE

Calculates finite element shape functions.

7.2.26 SOLVE

Solves banded, non-symmetrical matrices and stores the result in the force vector. Does not check for positive definiteness.

7.2.27 SWELL

Calculates gas released and swelling using calculated results.

7.3 Input

Input to program REDSTONE is done through 80 column character records residing in an external file. An example can be found in Appendix 2. Each record is first scanned for a '*' or 'C' in column one to see if the record is a comment. If so, parsing is stopped and the next line read in for processing. If the line is not a comment it is first parsed into words (denoted by a breaking space). The first word is checked to see if it is a valid identifying line number. An invalid line number causes program execution to stop with an error. Numbers are used in the first word of a line to denote the type of data present. The numbers used are grouped by hundreds with subgroups by tens. Thus a three (to four) digit number is required with the first two (or three) digits being significant. The last digit can be used at the user's discretion. Once the data type has been determined, a check is performed to insure the right amount of data is present on the line. If so, the individual words are assigned to the respective variables and the next line read in for processing. After processing all lines, the assigned data is printed to the output file. The specific types of data input are reviewed below by identifying line number.

7.3.1 Card Number 100

Contains user specified header information. This information is not parsed, but is printed as is on the output file. Only the '1' in the line number is significant, leaving numbers 100-199 to be used by the user for headers.

7.3.2 Card Number 200

Can contain a variety of data depending on the line number digit in the tens place. An identifier for whether British or metric units are used is identified by '210'. The number of Gauss points to use for integration is identified by '220'. A 0/1 flag for whether the input deck is reprinted on the output is denoted by '230'. A 0/1 flag to indicate whether a restart file is to be read is denoted by '240'. Currently, the type of units identifier is not used. Program input and output are in cgs units. Both the '2' and the second digit in the line number are significant.

7.3.3 Card Number 300

Contains the number of elements, the number of bubble sizes per node, the number of nodes per element, the number of specified concentrations, the number of time periods, and the maximum number of iterations to allow. Currently, the number of nodes per element is hard coded into much of the program for three nodes per element, so '3' should be the only value for this variable. The maximum number of iterations is applied individually for each iteration cycle, not as an aggregate for all iterations. Only the '3' in the line number is significant.

7.3.4 Card Number 400

Previously used to input the connectivity array of local node numbers to global node numbers. These are now calculated in the program and thus number 400 is no longer used.

7.3.5 Card Number 500

Contains the location of nodes in centimeters from the center of the fuel grain. A separate line is used for each node. Only the '5' in the line number is significant.

7.3.6 Card Number 600

Contains specified concentration information. Each specified concentration occupies one line. Each line contains a number identifying the bubble size, the node of the specified concentration, and the value of the concentration in cgs units. Only the '6' in the line number is significant.

7.3.7 Card Number 700

Contains the radius of the fuel grain in centimeters. Only the '7' in the line number is significant.

7.3.8 Card Number 800

Previously used to input material type identifiers for multiple material types (i.e., different enrichments, etc). Currently much of the program is hard coded for only one material type. As a result only one material type is allowed and this line number is no longer used.

7.3.9 Card Number 900

Contains the fuel porosity, enrichment, and convergence limit to use. The porosity and enrichment are fractions less than one. The porosity is not currently used since it is an intergranular property. Only the '9' in the line number is significant.

7.3.10 Card Number 1000

Contains the time length of periods in seconds, the number of time steps to take in a period, and the interval in time steps which to output results. One line is used for each time period. Only the '10' in the line number is significant.

7.3.11 Card Number 1100

Contains period operational data, one line for each period. Each line contains the initial power density in Watts per cubic centimeter, the initial temperature in degrees Kelvin, the initial temperature gradient in degrees Kelvin per centimeter, the initial pressure in Pascals, the final power density, the final temperature, the final temperature gradient, and the final pressure. Only the '11' in the line number is significant.

7.4 Output

Program REDSTONE produces several forms of output. The primary output is a listing of calculational results at user specified intervals. An example can be found in Appendix 3. The first part of this listing is a 'REDSTONE' banner followed by the user input header statements. This is optionally followed by a reprint of the input file. The input data is then summarized and printed in tabular form. Along with the input data are tables of a few preliminary calculations such as connectivity arrays. All the above information is printed only once. The rest of the output listing consists of the results of calculation at the end of time steps at intervals specified by the user. Each of the results listings begins with time step information, followed by the percent fission gas released and percent swelling. The calculated burnup in several forms is listed next followed by the power density, temperature, and pressure used. Finally, the bulk of the results listing involves the calculated bubble concentrations, radii, and pressures by node.

In addition to the above output, two other files containing the calculated results is written. One of these consists of only numbers separated by spaces and is suitable as input to other analysis programs. The data included in this file are the node number, the radius of the node, the bubble size, the bubble concentration, the bubble radius, and the bubble pressure on each line.

The second is a machine readable restart file which can only be read by program REDSTONE. This file is written automatically at the end of each time period and overwrites any previous information in the file. It contains the last calculated time period values only. This includes the convergence denominators, the total fission gas atoms produced, as well as the bubble, vacancy, and interstitial data, i.e. concentrations, radii, and pressures. As a check for restart, this file also contains the number of elements, number of bubble sizes, and number of nodes per element used. This information is compared with the input file upon restart to ensure that a compatible problem is being restarted. A non-compatibility results in an immediate program halt and an error.

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CHAPTER VIII RESULTS

8.1 Validation of Program REDSTONE Internals

Program results will never be valid if the program doesn't perform as intended. So that the internal consistency of REDSTONE could be checked, calculated results were compared with analytic solutions where possible. At least two analytic solutions are available for comparison. Both are linear, single bubble size solutions. The first is the steady-state solution and the second is a time dependent solution for simple diffusion with a constant source.

8.1.1 Steady-State Solution

For a single bubble group in 1-D spherical geometry, the steady-state equation is

$$\frac{D}{r^2}\frac{\mathrm{d}}{\mathrm{d}r}\left(r^2\frac{\mathrm{d}C}{\mathrm{d}r}\right) + K = 0 \tag{198}$$

where D = diffusion coefficient, K = production source, and C = concentration. The boundary conditions are

$$\frac{\mathrm{d}C}{\mathrm{d}r} = 0 \quad \text{at} \quad r = 0 \tag{199}$$

and

$$C = 0 \text{ at } r = R_{g} \tag{200}$$

where R_g is the grain radius. The analytic solution to this equation is

$$C = \frac{K}{6D} \left(R_{\rm g}^2 - r^2 \right) \,. \tag{201}$$

The constants K, D, and R_g can be defined with reasonable values, i.e. K=4.104 ×10¹¹ s⁻¹cm⁻³, $D = 1.38 \times 10^{-18}$ cm²s⁻¹, and $R_g = 10 \mu$ m. Using these values in program REDSTONE and in the analytic solution results in the comparison shown in fig. 3. Exact agreement is obtained by REDSTONE as would be expected for a strictly linear smoothly varying solution.

8.1.2 Time-Dependent Solution

The problem that was solved above was simple. A more difficult problem involves the time dependence since this will introduce near discontinuities in the slope of the solution at the grain boundary. The time dependent equation for a single bubble group in 1-D spherical geometry is

$$\frac{\partial C}{\partial t} = \frac{D}{r^2} \frac{\partial}{\partial r} \left(r^2 \frac{\partial C}{\partial r} \right) + K \quad (202)$$

The same boundary conditions are used as before with the added initial condition of C = 0 at t=0. For these conditions, a series solution can be obtained [6], i.e.

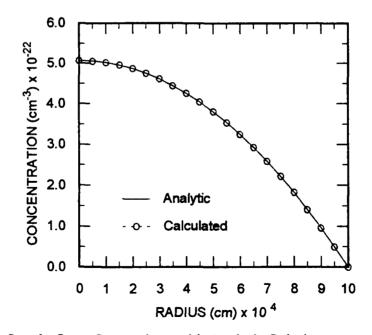


Fig. 3 Linear Steady-State Comparison with Analytic Solution

$$C = \frac{K}{6D} \left(R_{\rm g}^2 - r^2 \right) + \frac{2KR_{\rm g}^3}{D\pi^3 r} \sum_{\rm n=0}^{\infty} \frac{(-1)^{\rm n}}{{\rm n}^3} \sin\left(\frac{n\pi r}{R_{\rm g}}\right) \exp\left(-\frac{D{\rm n}^2\pi^2 t}{R_{\rm g}^2}\right) .$$
(203)

Calculations using REDSTONE for time = 3×10^5 secs are compared with the series solution in fig. 4. Although the REDSTONE calculation represents the series solution well near the center of the grain, it oscillates wildly at the outside of the grain. This oscillation is caused by the inability of the interpolation functions to adequately represent the behavior of the concentration in a region of sharp change and is similar to shock behavior in mechanical modeling. Ideally, one would want interpolation functions which followed the profile of the concentration exactly. This is not generally possible for real world problems. However, one can improve the approximation of the concentration profile in several ways, two which are used here.

The first improvement involves adjusting the interpolation functions in the outermost element to better approximate the sharp decline in concentration at the grain boundary. Burnett [120] notes that the slope of three node interpolation

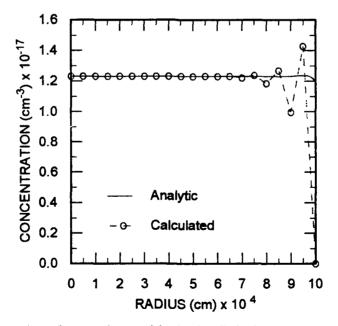


Fig. 4 Linear Transient Comparison with Series Solution

functions at one edge of an element can be made infinite by moving the central element off center to a position 0.75 of the distance between the two outer nodes. This arrangement and the effect on the interpolation functions are shown in fig. 5. The effect on the results from REDSTONE are compared with the original REDSTONE calculation in fig. 6. A noticeable lessening of oscillatory behavior has occurred. However, oscillations remain because the interpolation functions still do not exactly represent the concentration.

A second way to improve the representation is to increase the number of elements and nodes in the vicinity of the change. This was tried for REDSTONE using the element scheme as outlined in fig. 7. Although the same number of elements as used previously are present, their spacings have changed so that much

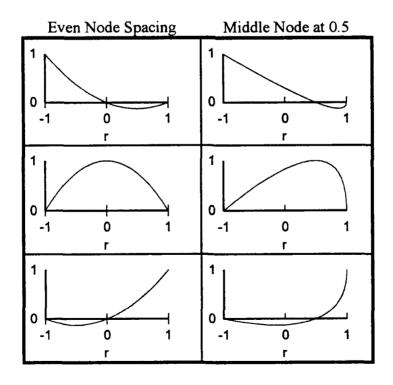


Fig. 5 Change in Interpolation Functions with Middle Node Shift

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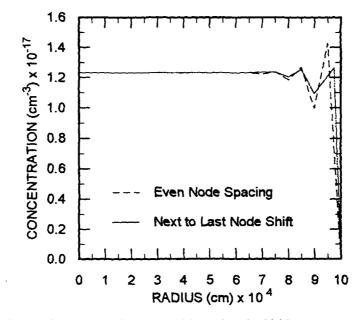


Fig. 6 Comparison of Results with and without Node Shift

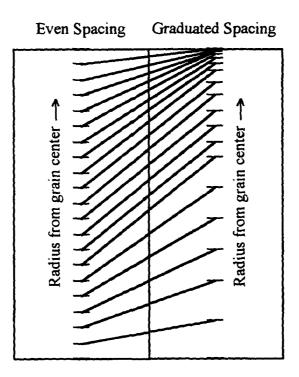


Fig. 7 Graduated Node Density Spacing in Fuel Grain

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more detail can be modeled near the grain boundary. The elements sizes gradually diminished from inside the grain to the surface. The REDSTONE calculation using this elemental scheme (and the node shift) is compared to the previous REDSTONE calculations in fig. 8. Oscillatory behavior has now become nearly negligible, being reduced by a factor of at least 1000. Increasing the nodal density in the region of change has caused a significant improvement in REDSTONE's ability to model actual distribution profiles.

It is apparent from the above analysis that a combination of correct finite element programing and an intelligent approach to the problem are required to achieve acceptable results. Through the methods presented above, REDSTONE has been shown to be capable of adequately representing the problem at hand.

8.2 Validation of Program REDSTONE Results

A program may work well yet still provide unacceptable results if the correct equations are not programmed. Comparison with known experimental results provides indications of how well a program models the real world. Those comparisons for REDSTONE are presented here.

8.2.1 Additional Constraints

Initial calculations of multi-bubble problems utilized the full radius dependent formulations of bubble radii change presented in Chapter VII. It was quickly realized that the bubble radii calculations were extremely sensitive to changes in the radii. If bubble radii changed too much in a given time step, the program would begin to diverge. Substitution of the simpler partial-radius dependent formulation improved the stability of the program by slowing the rate of change. However, the bubble radii calculation was still very sensitive to the size of

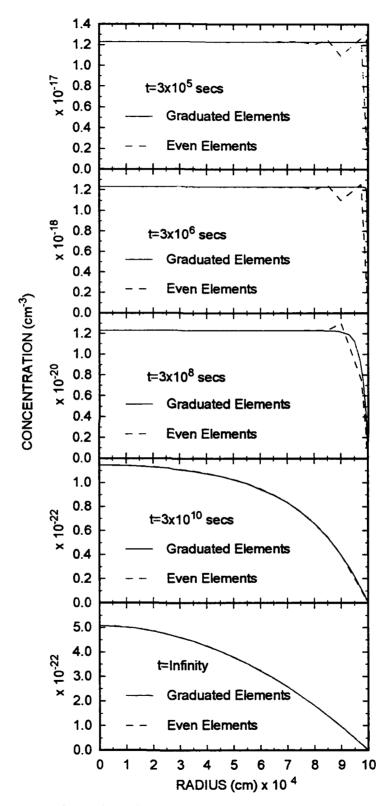


Fig. 8 Comparison of Results With and Without Graduated Node Density

change. A contributing factor to program instability was found in the calculation of bubble pressures. Due to the form of the reduced van der Waals equation of state, it was possible to calculate negative pressures if the radii became too small. For the small highly pressurized bubbles here, this occasionally happened in the course of calculations. These negative pressures affected calculations through the vacancy emission terms in both vacancy concentration and radii calculation. The bubble pressure was found to be very sensitive to bubble radius in this region and the overly-small radii were always only a few percent away from a radius that would produce a positive pressure. This problem was solved by limiting the pressure of a bubble to a maximum pressure. The yield pressure is a natural choice for this maximum pressure. Unfortunately, the yield pressure of UN as measured in the laboratory exhibits strain hardening, making if difficult to ascertain what the appropriate value to use within the grain is. Because of this a crude approximation of the yield pressure was made by using the curves of Werner and Blank [84], as mentioned in Chapter III. The hope was that these curves could at least provide a temperature dependence and an indication of the pressure needed for plastic deformation. Fortunately, the bubble pressure is very sensitive to radius in this region. Conversely, the radius is insensitive to pressure. Thus large changes in pressure do not significantly affect the radius in the region near the yield point. A few preliminary calculations using REDSTONE proved this to be the case. Thus the correlation presented in Chapter III was successfully used to eliminate problems with negative calculated pressures.

This, however, did not eliminate the sensitivity of the radii calculations to changes in bubble radii, especially during the severe transient to be modeled here. It was found that the vacancy emission term was still the destabilizing term. The unique formulation of this term due to the pressure dependence may have caused a loss of too much of the non-linearities. This could have then resulted in the inability of the correction term to drive the solution in the correct direction when changes became too large. It was found that the calculations could be stabilized by limiting the maximum size of the bubbles to the equilibrium size. This was an acceptable constraint since bubbles would be expected to remain at or below their equilibrium size in steady-state or ramp-up conditions. However, in a ramp-down condition, bubbles are likely to become underpressurized resulting in sizes larger than the equilibrium size. Enforcing a maximum bubble size constraint in this situation causes the bubbles to attain their equilibrium size which might affect the calculated results.

8.2.2 Comparison with Experimental Studies

With program calculations stabilized, REDSTONE was ready to compare with experimental studies. Unfortunately, the number of experimental studies using UN are a small group. In addition, those studies measuring the types of data needed for comparison to intragranular code predictions are even fewer. No studies of UN were found where a microscopic determination of swelling components were made. Studies which were found included only those measuring macroscopic swelling and gas release. Most of these studies, especially the more recent ones, have dealt with fuel temperatures much higher than the those of interest here. This further reduces the candidates for comparison since program REDSTONE has the added restriction that the species modeled cannot be too mobile. Large bubbles are produced when the species are mobile enough for significant coalescence to occur, as with high temperatures. This is a restriction because only individual bubble sizes are modeled rather than bubble groups with averaged properties. As a result, large bubbles can only be modeled using very large program arrays, which take up large amounts of computer storage and exponentially increase the running times. Local computer constraints restricted the size of the problem here to 50 bubble sizes. Even at 50 bubble sizes, the running times increased over six-fold when the number of bubble sizes was doubled. Thus, only studies utilizing lower temperatures, shorter burnups, or both were candidates for comparison.

When comparing REDSTONE results to experimental release and swelling studies, one should keep in mind that these studies are based on release from fuel pellets, not individual fuel grains. In general, since gas released from the grain will be collected on grain boundaries before release from the pellet, the release reported in the studies should be lower than that predicted by an intragranular code. However, some minimum amount of release would be expected from secondary knockouts and recoil atoms on the pellet surface and in any open porosity. Direct recoil release will come from fissions within approximately 10µm of the pellet surface, while knock-ons, with less energy, will only come from about 200 Å deep into the pellet [5]. Olander derives theoretical equations for release rates for both these mechanisms. After noting that the equations only predict a 6% release from 10% burnup fuel in a 0.6 diameter pellet, he concludes that these mechanisms are not very important in reactor fuel-element performance. However, this calculated release is equivalent to 0.6% release per atom% burnup, which although low, is significant when comparisons are to be made. While Olander's result is only a theoretical release rate, it does indicate that very low, but measurable, releases of fission gas from fuel could be explained entirely by recoil and knockout mechanisms.

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With regards to swelling, it needs to be noted that swelling at the grain boundaries and from non-gaseous fission products are not included in REDSTONE's calculations. Olander [5] has stated that this additional volume swelling amounts to between 0.15 and 0.45% per atom% burnup for UO_2 , depending on the stoichiometry of the fuel. A similar range can be assumed for UN. Additionally, any significant release to the grain boundary will cause larger swelling rates than in the interior of the grain. Thus, swelling predictions by REDSTONE should be less than in experimental studies, with better agreement obtained for low grain boundary release situations.

Two studies were found which satisfied the lower temperature, lower burnup criteria. One was a swelling study performed at the National Aeronautical and Space Administration (NASA) in 1973 [121]. The second was a release and swelling study performed at the Japan Atomic Energy Research Institute (JAERI) in 1977 [122]. In addition, Storms [123] has correlated experimental release data to theoretical density, burnup, and temperature for releases between 0.1 and 15%. He compared his correlation with similar correlations by Ross [124], Baars [125], and Thomas etal [126]. He notes that his correlation, like others, tends to underpredict release for large releases. Although he does not discuss it, his correlation also overpredicts release for low burnup, low temperature situations as is the case here. This is also true for the correlation by Thomas et al, but not true for those by Ross and Baars. Partly this is because there is a large scatter in the data, and the various sets of data used by each were slightly different. However, Storms also discarded from his data all releases below 0.1% regardless of irradiation temperature. This eliminated most of the low temperature, low burnup data, i.e 85% of the data below 1300°K / 3% burnup, and 70% of the data below 1500°K / 3% burnup. Additionally, he includes

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data from U-Pu-N, as does Thomas et al., which typically showed much larger icicases at the same temperatures and burnups. As a result, these latter two correlations are skewed to predict larger releases than experiment has shown in this low temperature, low burnup region. It should be noted, however, that the amount of error introduced by these correlations is not generally important since the releases are low. In consideration of the above, rather than use a correlation to compare with REDSTONE results, the data tabulated by Storms will be used to indicate the range of release expected experimentally for each of the studies below.

8.2.3 Comparison with NASA Results

The study at NASA measured swelling of six UN fuel pins containing four pellets each which were irradiated for approximately 4000 hours to burnups of up to two percent. The average cladding surface temperature was 1100°K. However, there was some degree of thermal cycling due to reactor reloads. Swelling data was obtained by removing three pellets from two of the pins for measurement. Of these, one had significantly lower burnup which allowed the results to be compared with REDSTONE calculations. Reported fuel characteristics and operational conditions for this pellet are given in table 5.

The NASA swelling results are compared with the REDSTONE result in table 6. Also in table 6 are REDSTONE release data and Storms' data for burnups between 0.74 and 1.36% and temperatures below 1400°K. In performing the REDSTONE calculation, the fuel grain size and exact operating history were not known. The grain size used was 20µm diameter and no thermal cycling was considered.

Table 5

Fuel Pellet Characteristics of NASA Study

Table 6

Comparison	of NASA	A Results with REDSTONE Results

	$\frac{\sqrt{2}}{2} \Delta V/V$	% Release	
NASA	0.99 - 1.8		· · · · ·
Storms		0.01-0.10	
REDSTONE	0.367	0.0567	

The values obtained by REDSTONE are lower than other values in table 6. Some of this difference might be due to the thermal cycling in the operating history. Conceivably, the thermal cycling could have involved short periods of higher temperature causing greater coalescence of bubbles, greater release, or intra-pellet cracking which could have reduced the overall density of the pellet and conversely increased volume. Additionally, the chemistry of the fuel pellets was not controlled to ensure that stoichiometric fuel was used. It was previously mentioned that other investigators were having trouble with non-stoichiometry around this period, which would have resulted in increased diffusional properties [123].

Some additional swelling was almost certainly contributed by non-gaseous fission products. When this additional swelling is considered, program REDSTONE's swelling result falls close to the range experimentally observed. The residual difference can be attributed to the factors mentioned above.

REDSTONE's gas release result falls in the middle of the data tabulated by Storms. One would expect release from the grain to exceed that from the fuel pellet, but in fact a small intragranular release may not be released outside the fuel pellet at all. It is likely that the smallest release results tabulated by Storms involved release by the recoil and knockout mechanisms, not by release from grains in the interior of the fuel.

8.2.4 Comparison with JAERI Results

The investigators at JAERI studied carbide, carbonitride, and nitride fuels irradiated to burnups of 24000 MWD/t at temperatures between 1264°K and 1634°K. Controllers were used to maintain temperatures within \pm 50°K during irradiation. Three UN pellets were measured for both gas release and swelling. Two of these had low enough burnups to be compared with REDSTONE results. Reported fuel characteristics and operational conditions for these pellets are given in table 7.

The JAERI swelling results are compared with the REDSTONE results in table 8. Also in table 8 are Storms' data for burnups equivalent to that in the JAERI study, between 1.36 and 2.34% and temperatures below 1400°K. In performing the REDSTONE calculation, the fuel grain size used was 20µm diameter.

As with the NASA study, the REDSTONE results are on the lower end of the swelling range observed experimentally, although slightly better agreement is obtained. When non-gaseous swelling is considered the REDSTONE results fall in the middle of the observed swelling ranges. However, experimental swelling data can be seen to vary widely. This variation, besides making comparisons with REDSTONE results tenuous, indicates that in low fission gas swelling situations,

Table 7Fuel Pellet Characteristics of JAERI Study

	Pellet 1	Pellet 2	
Outside Diameter (cm)	0.325	0.345	
Length (cm)	0.7	0.7	
Enrichment (%)	20	5	
Theoretical Density (%)	88-95	88-95	
Power Density (W/cm ³)	1998	2224	
Average Temperature (°K)	1169	1344	
Burnup (MWD/t)	12000	16000	-

Table 8

	% ΔV/V	% Release	% ΔV/V	% Release
JAERI (Pellet 1)	0.5-1.8	1.2		
JAERI (Pellet 2)	***		0.7-2.4	0.7
Storms		0.09-1.91		0.09-1.91
REDSTONE	0.4619	0.0223	0.7200	0.0185

other processes are contributing as much or more to the overall swelling of the fuel. Some of the possible processes were discussed previously. Based on the comparisons made here, REDSTONE appears to be satisfactorily accounting for the intragranular fission gas component in the overall swelling. In general, however, it appears that REDSTONE results will be below experimentally observed swelling percentages by as much as 1.5%.

Fission gas release results are also compared in table 8, with REDSTONE results being below that of the JAERI study but within the experimentally observed range of Storms. Because the percentage release is low, the same arguments apply here as for the NASA study. Most experimentally measured release likely did not come from diffusion out of fuel grains.

8.3 Modeling of the Transient Scenario

The transient scenario of interest here involves sustained low temperature operation followed by a rapid surge in power by at least two orders of magnitude which is sustained for several minutes. The parameters of this scenario were discussed earlier. Table 9 lists those parameters which were used in REDSTONE to model the scenario.

Results of the REDSTONE calculation are tabulated in table 10. These results show that the swelling before the transient is small and that the transient has little effect on the overall swelling and release, presumably because the time at high temperature is short. The amount of gas released is insignificant, especially when release from other non-diffusional sources are considered. The post-transient swelling of 1.0649% corresponds to a diametral increase of 0.354%. If one adds to this the possible 1.5% swelling due to non-gaseous processes, the diametral increase is still only 0.848%. A check of the data in table 1 will reveal that the fuel must increase in diameter 4.6% before coming into contact with the liner, and assuming unrestrained swelling against the cladding, must swell 5.9% before causing design failure. Design failure was specified as 1% diametral increase of the fuel pin. Alternately, swelling of the fuel pin because of released gas is not indicated since the gas release from the grain is insignificant. It is clear that the results reported here do not indicate fuel pin failure will occur due to swelling or release of fission gases from the fuel.

Specific characteristics of the solution are illustrated in figs. 9-12. Fig. 9 shows the swelling and gas release during the transient and high temperature periods. As seen in fig. 9, the swelling initially increases with the transient, lagging slightly, then continues to increase. The percentage release shows no effect from the

Table 9

Transient Scenario Modeling Parameters		
Fuel Characteristics		
Fuel	UN	
Theoretical density (%)	94.3	
Enrichment (%)	93	
Grain diameter (µm)	_ 20	
Operational Characteristics		
Initial power (W/cm ³)	52.6	(equivalent to 10 kW electric)
Initial temperature (K)	500	
Initial temperature gradient (K/cm)	38	
Time at initial power (sec)	3×10^{8}	
Transient time (sec)	30	
Final power (W/cm ³)	5260	(equivalent to 1 MW electric)
Final temperature (K)	1500	
Final temperature gradient (K/cm)	2550	
Time at final power (sec)	1000	-

Table 10

Transient Scenario REDSTONE Results

	Pre-transient	Post-transient			
Volume Swelling (%)	1.0627	1.0649			
Gas Release (%)	*	5.274×10 ⁻³			
* Pre-transient gas release was below the significance maintained by the computer.					

transient and continues to decrease slightly. Not shown is the pre-transient time profile, but this would have shown an initial percentage release of gas which reaches a maximum and then begins a steady decrease once the bubble concentration has been built up enough to prevent significant diffusional loss from the grain. The bubbles do not contribute to any significant release, either before the transient or after.

An inspection of the characteristics shown in the other figures indicates that the swelling rate during the high temperature region, although small, is greater than that from simple deposition of gas atoms in the lattice. Fig. 10 shows the concentration profile in the center of the grain both immediately before the transient

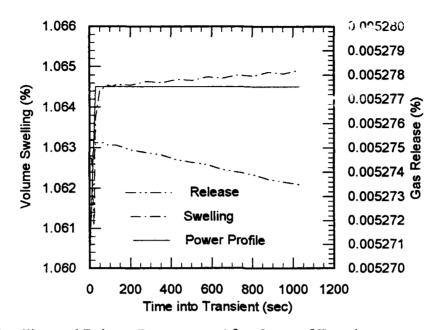


Fig. 9 Swelling and Release Percentages After Onset of Transient

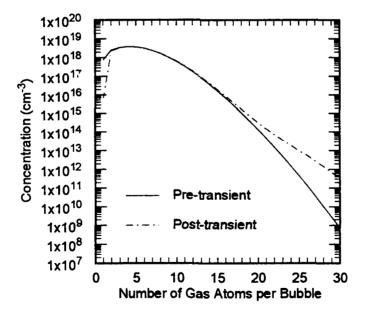


Fig. 10 Pre- and Post-transient Concentration Profiles

and at the end of the high temperature period. The concentration profile significantly changes only for single gas atoms and in the large bubble tail of the profile. The single gas atom concentration is becoming depleted, and the larger bubbles are beginning to grow in number. The highest concentration of bubbles are those containing four gas atoms both before and after the transient. It is clear that there is a shift occurring to larger bubbles. The question remains, however, on whether this translates into an increase in swelling. A check of the bubble radii shown in fig. 11, both before and after the transient, shows little change occurring, with only a slight increase for larger bubbles. This is supported by fig. 12 which shows the change in bubble pressures. The smaller bubbles have actually increased in pressure after the transient, as would be expected if the temperature rose but the radii remained relatively constant. The largest bubbles, on the contrary, started out overpressurized and actually decreased in pressure after the transient. This could only happen if the radii had become larger. The implications of the above analysis

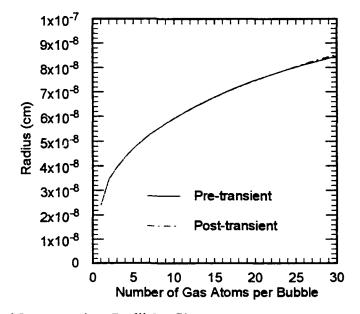


Fig. 11 Pre- and Post-transient Radii Profiles

is that swelling is increasing faster than the deposition of new gas atoms. While the changes in swelling are extremely small and not important for the scenario modeled here, they do indicate processes occurring which may become of importance at higher temperatures. To investigate this possibility, the scenario above was modeled at higher temperatures. This investigation is discussed below.

8.4 Variation of Fuel Temperature

An investigation into varying the fuel temperature was done by increasing the pre- and post-transient temperatures by 100°K, 200°K, and 300°K. Thus the pretransient temperatures modeled were 500°K, 600°K, 700°K, and 800°K. The corresponding post-transient temperatures were 1500°K, 1600°K, 1700°K, and 1800°K. As shown in fig. 13, the swelling increases steadily with temperature both before and after the transient. However, the amount of increase is small and not important for the scenario modeled.

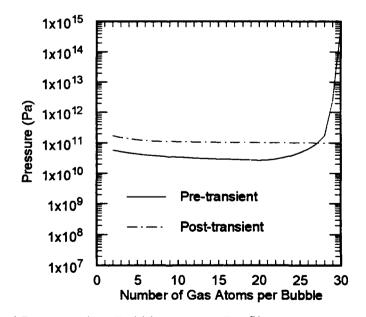


Fig. 12 Pre- and Post-transient Bubble Pressure Profiles

The associated gas release presented in fig. 14 shows only post-transient release as the pre-transient release was extremely small and essentially unaffected by the temperature. This lack of pre-transient temperature effect is supported by the pre-transient concentration profiles in fig. 15 which show virtually no change in profiles with temperature. The post-transient release also changes little with temperature except for the results at 1800°K. The small changes in release at 1500°K, 1600°K, and 1700°K indicates that bubbles still are not significantly contributing to release, and in fact are keeping the single gas atoms from being released. The larger release at 1800°K may indicate that the bubbles are becoming more mobile at this temperature.

A check of the post-transient concentration profile in fig. 16 supports this assertion. In fact, the concentration profiles at all temperatures show marked

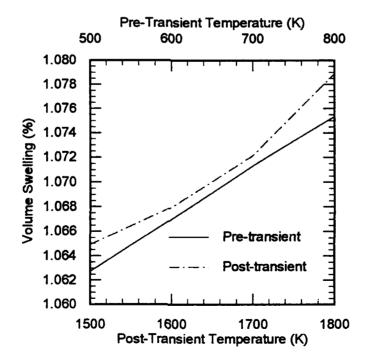


Fig. 13 Volume Swelling vs. Pre- and Post-transient Temperatures

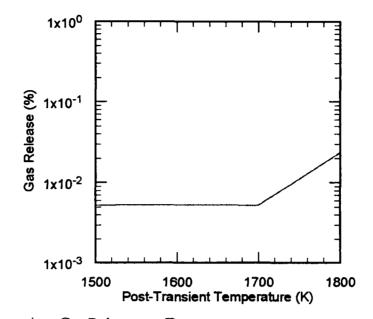


Fig. 14 Post-transient Gas Release vs. Temperature

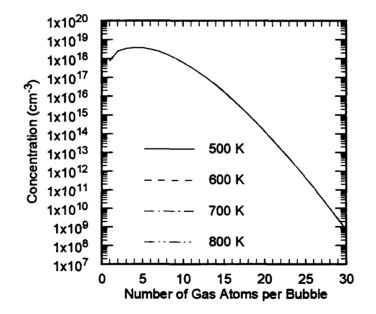


Fig. 15 Pre-transient Concentration Profiles vs. Temperature

differences between each other. The concentration of the largest bubble size increases significantly with temperature until, at 1800°K, it appears to be on the leading edge of an upward trend and is of the same order of magnitude as the highest concentration shown. Additionally, the profiles at 1700°K and 1800°K show oscillations at the smaller bubble sizes. This is possibly due to the increasingly depleted concentration of single gas atoms with higher temperature which induces a calculational shock to the solution. Increasing the number of time steps in the transient should reduce this oscillatory behavior.

Investigating the concentration profiles at the higher temperatures further, it appears that the bubbles are becoming too mobile for REDSTONE to be reliable at 1800°K and possibly at 1700°K. Fig. 16 would indicate that there are much larger bubbles at the same or higher concentrations at 1800°K. These larger bubbles have the potential to significantly increase the actual swelling over what REDSTONE has calculated. Even when the 1800°K scenario was run with 50 bubble sizes, the

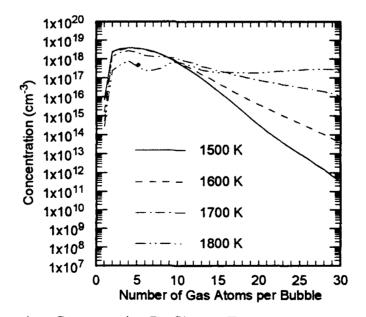


Fig. 16 Post-transient Concentration Profiles vs. Temperature

profile remained relatively flat. Without a reasonable indication that nearly all the bubble sizes are accounted for, it is questionable to rely on REDSTONE's results.

So that some measurable indication could be used, a figure-of-merit was developed to help discriminate when REDSTONE's results might be unreliable. This figure-of-merit was simply a ratio of the concentration at the maximum in the profile to the concentration in the largest bubble size. This is plotted in fig. 17. The effect of increasing temperature on post-transient results can clearly be seen in fig. 17. The question now is where to draw the line below which results are considered to be unreliable.

For the purposes here, a liberal definition of a reliable REDSTONE result was chosen. If the figure-of-merit was above 100, the result was considered reliable. Using this criteria, the results at 1800°K would clearly be unreliable, while those at

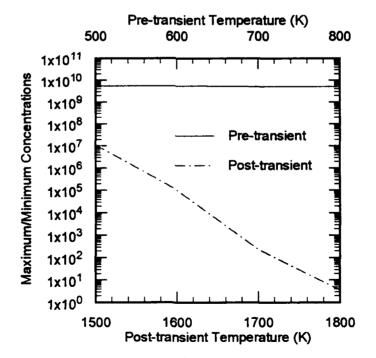


Fig. 17 Pre- and Post-transient Figures-of-Merit

1700°K, while close, would be considered reliable. The previously mentioned modeling of 50 bubble sizes at 1800°K resulted in an equally poor figure-of-merit as with 30 bubble sizes.

The most important observation of the above $c^2 = 3$ sion is that the swelling and release behavior at 1800°K remains unknown. Indications from REDSTONE are that bubble mobility becomes appreciable at this temperature, leading to large bubble sizes and possibly significant swelling. This is important since scoping calculations indicated a fuel centerline temperature of around 1800°K, even though the average temperature of the fuel would remain cool enough that swelling would not be a problem. However, the calculations here assume a perfectly uniform system. In an operating reactor, fuel will vary in temperature by position in the core and possible imperfections in the fuel or cladding. Thus, some pellets might have average temperatures near or above 1800°K, and the temperatures would certainly increase if final power were boosted with minimal changes in design. Because this temperature is close to the temperatures expected to be present during the current transient scenario, knowing the gas behavior at 1800°K is an important design prerequisite. Unfortunately, REDSTONE is not readily capable of producing this without some sort of bubble grouping scheme. Additionally, in consideration of the large uncertainties in some of the input properties used in REDSTONE, one should not discount the possibility that increased bubble mobility actually becomes a problem at lower temperatures than predicted by REDSTONE.

8.5 Variation of Property Data

Sensitivity studies were performed to determine whether the uncertainties in properties could cause a significant effect on the results. The properties investigated

included the xenon, surface, and irradiation enhanced diffusion coefficients, the surface energy, and the resolution parameter. The same transient scenario investigated above was used except that a much coarser time stepping routine was used to speed program operation.

Each of the following discussions utilize a gas release plot showing posttransient gas release, a pre- and post-transient swelling plot, and the figure-of-merit plot discussed previously. Except for surface energy, the properties were varied through scaling by factors of 0.1, 1, 10, and 100. The resolution parameter was additionally scaled by 0.01. Surface energy was compared at 780, 1000, 1500, and 2000 ergs/cm².

8.5.1 Xenon Diffusion Coefficient

The gas release results obtained by scaling the xenon diffusion coefficient are presented in fig. 18. As expected, an increase in post-transient release is seen

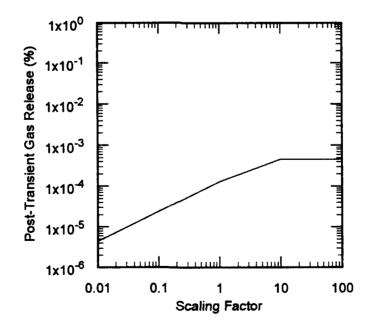


Fig. 18 Scaled Xenon Diffusion Coefficient Gas Release

with increased xenon diffusion coefficients. The reason for the leveling off for the highest diffusivities is unknown. Conceivably, this may have resulted from the quick depletion of gas atoms through release or capture.

The swelling results (fig. 19) show similar trends. As expected, the pre-transient swelling remains completely unaffected by the xenon diffusion coefficient since irradiation enhanced diffusion is by far the dominant process at the low pre-transient temperature. The post-transient results show the expected increase in swelling as increased gas atom capture occurs, but with an unexplained dip in swelling for the highest diffusivity. This dip corresponds with the leveling off in release results. A check of the figure-of-merits in fig. 20 indicates that all results should be reliable.

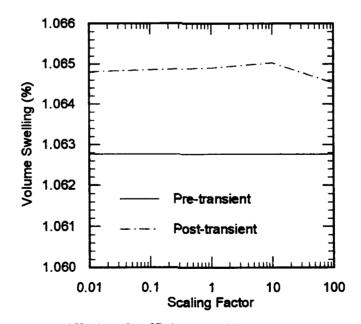


Fig. 19 Scaled Xenon Diffusion Coefficient Swelling

8.5.2 Surface Diffusion Coefficient

The gas release results obtained by scaling the surface diffusion coefficient are presented in fig. 21. Since bubbles do not appear to contribute to gas release, it was expected that the surface diffusion would have minimal effect on the gas release. This is generally true except for the highest diffusivity. At 100 times the surface diffusion, it appears that bubbles began contributing to the gas release. Again, however, the amount released is extremely small.

Because the release is small, one would expect that increased surface diffusion would result in more coalescences leading to higher concentrations of large bubbles and higher swelling. Indeed, this is seen to be the case in fig. 22. Again, because of the already very small mobilities at the low pre-transient temperature, the pre-transient bubble motion appears unaffected by increased surface diffusion. A check of the figure-of-merit (fig. 23) supports this assertion and

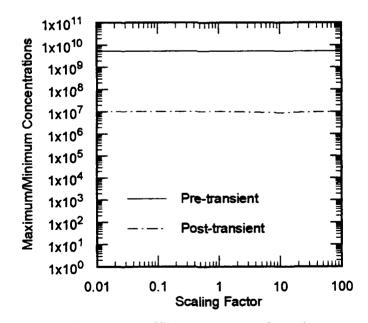


Fig. 20 Scaled Xenon Diffusion Coefficient Figures-of-Merit

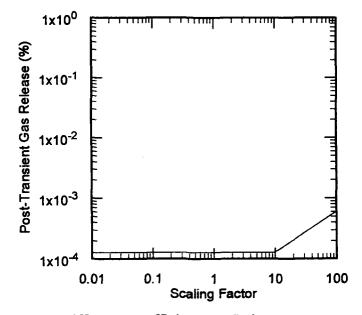


Fig. 21 Scaled Surface Diffusion Coefficient Gas Release

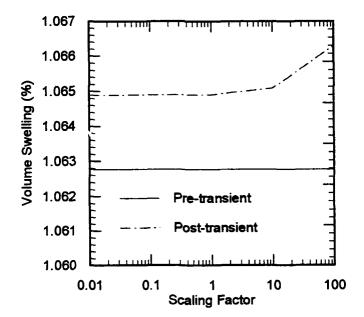


Fig. 22 Scaled Surface Diffusion Coefficient Swelling

the assertion that more coalescences are occurring. However, the figure-of-merit for the post-transient swelling at the highest diffusivity comes close to the reliability cutoff.

8.5.3 Irradiation Enhanced Diffusion Coefficient

The gas release results obtained by scaling the irradiation enhanced diffusion coefficient are presented in fig. 24. Although irradiation enhanced diffusion is important in the pre-transient region, release results in this region remained below the level of significance stored in the computer for all scalings of the diffusion coefficient. When the effect on swelling (fig. 25) is considered, it is seen that swelling increases with increased irradiation enhanced diffusion during the pre-transient period. This trend is the result of increased gas atom capture rather than release. The post-transient release results suggest this same trend since greater gas atom capture in the pre-transient period would result in a reduced gas concentration

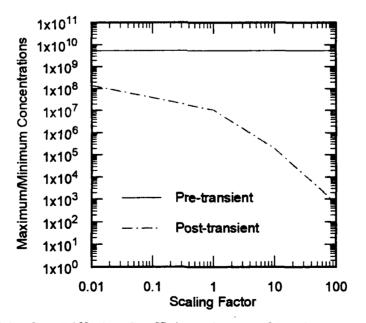


Fig. 23 Scaled Surface Diffusion Coefficient Figures-of-Merit

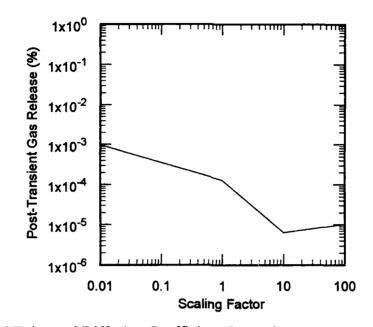


Fig. 24 Scaled Enhanced Diffusion Coefficient Gas Release

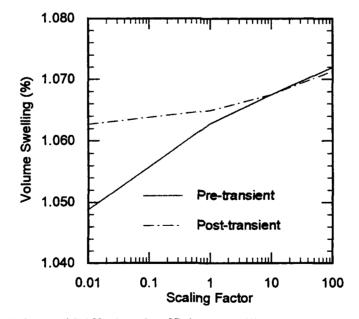


Fig. 25 Scaled Enhanced Diffusion Coefficient Swelling

being available for release after the transient. The bubbles do not appear to contribute significantly to release.

The swelling results shown in fig. 25 indicate both pre- and post-transient swelling increases with increasing irradiation enhanced diffusion. The pre-transient trend was discussed above. The post-transient swelling increases because of the corresponding increase in the pre-transient period. Interestingly, however, the post-transient results exhibit smaller increases compared to the pre-transient increases. If bubbles did not have time to move appreciably during the high temperature phase, one might expect a closer relationship between the two trends. Since they appear different, this could indicate that the post-transient swelling is only partially dependent on the pre-transient condition, and that coalescences after the transient play a significant role in setting up the post-transient profile. However, a check of the figures-of-merit (fig. 26) before and after the transient indicate that the results at he highest diffusivity are near the reliability limit, so there may be

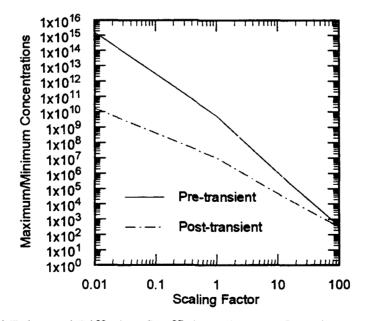


Fig. 26 Scaled Enhanced Diffusion Coefficient Figures-of-Merit

some uncertainty in the exact values calculated at 100 times the diffusion coefficient..

8.5.4 Surface Energy

The gas release results obtained by setting the surface energy at various energies are presented in fig. 27. Since the surface energy's main effect is in the determination of the size of the bubbles, it was not expected that gas release from the grain would be affected by changing the surface energy. This is demonstrated in the release results obtained.

The swelling results (fig. 28) clearly show the effect of changing the bubble sizes. As expected, increasing the surface energy decreases the swelling, and decreasing the surface energy increases swelling. There does not appear to be any significant effect on bubble mobility since the two curves in fig. 28 exhibit similar trends. A check of the figures-of-merit (fig. 29) indicates that all data are reliable.

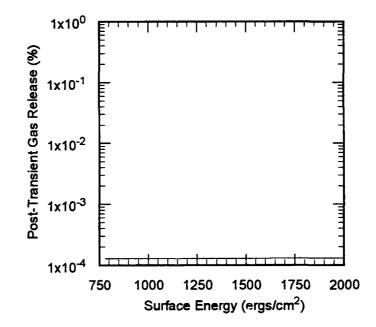


Fig. 27 Surface Energy Gas Release

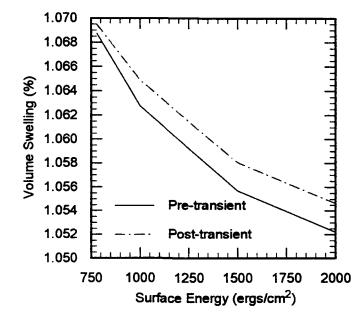


Fig. 28 Surface Energy Swelling

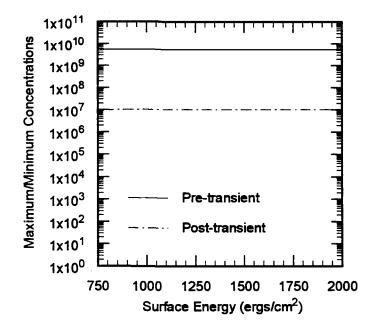


Fig. 29 Surface Energy Figures-of-Merit

8.5.5 Resolution Parameter

The gas release results obtained by scaling the resolution parameter are presented in fig. 30. A significant effect on release can be seen. Since the resolution parameter used here is one to two orders of magnitude lower than experimentally observed for UO_2 , these results show that UO_2 equivalent resolution results in a four order of magnitude increase in release. The amount of release is still not large enough to be of importance. The cause of the increase is easily understood. As resolution increases, more gas atoms are kept free in the lattice. These single gas atoms are then more likely to diffuse out of the grain than be captured and kept by a bubble.

A look at the swelling results in fig. 31 show interesting trends. The pretransient swelling exhibits a maximum at a scaling factor of one with decreased swelling as one either increases or decreases the resolution rate. The decrease in

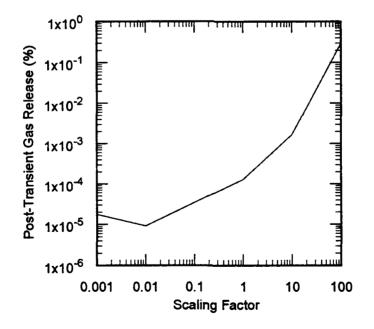


Fig. 30 Scaled Resolution Parameter Gas Release

swelling with increasing resolution is easily understood. The decrease in swelling with decreasing resolution is confusing until one checks the associated figures-of-merit in fig. 32. From these, it is clear that the concentration profile is flattening as resolution decreases. The lowest resolution results are clearly unreliable, and the next-to-lowest results are suspect. The pre-transient result at 0.1 cannot be eliminated based on the criteria that has been set up. The reason for this decrease is not clear. Since the bubbles are essentially frozen at the low pre-transient temperature and thus bubble concentration grows primarily through gas atom capture, one could argue for some decrease in swelling with lower resolution if the single gas concentration became so low that few were left to adequately develop a significant concentration of large bubbles. However, this argument seems tenuous, at best.

Considering, now, the post-transient swelling one sees that it exhibits a much milder dependence on resolution. As expected, it decreases with increasing

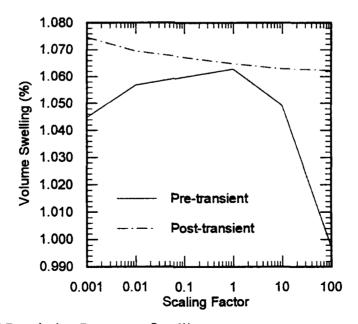


Fig. 31 Scaled Resolution Parameter Swelling

resolution rate. However, it appears that this is further evidence that post-transient mobilities may have had an effect in setting up the post-transient concentration profile.

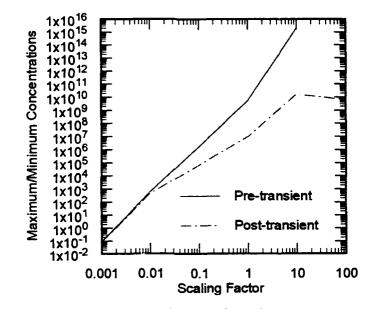


Fig. 32 Scaled Resolution Parameter Figures-of-Merit

CHAPTER IX SUMMARY AND CONCLUSIONS

9.1 Summary

In order to investigate whether fission gas swelling and release would be significant factors in a space based nuclear reactor operating under the Strategic Defense Initiative (SDI) program, a finite element program was developed to model the transient fission gas behavior in a single uranium nitride fuel grain. Only the intragranular problem was investigated because of the importance of the intragranular fission gas as a precursor to all other fission gas based processes, and because it was necessary to limit the investigation to a reasonably sized problem.

The program developed was named REDSTONE (Routine for Evaluating Dynamic Swelling in Transient Qperational Nuclear Environments) and was based on formulation of the governing 1-D, spherical geometry diffusion equations using the Galerkin method. The equations were incrementalized so that non-linear dependences would be retained, and the solution was obtained through iteration. Calculations were divided into three sets of governing equations: bubble concentration changes, bulk vacancy and interstitial concentration changes, and bubble radii changes. The bubble concentration and radii equations were derived from considerations of production and loss mechanisms of individual bubble groups defined by the number of gas atoms in the bubble. This is in contrast to similar investigations for UO_2 which have utilized groups with average properties describing bubbles containing a range of gas atoms. The choice to model individual bubbles placed limitations on the types of scenarios REDSTONE could model. The limitation was due to the increased storage requirements and running times needed

to model a large range of bubble groups. Since higher temperatures and longer burnups cause larger bubbles to form, REDSTONE's calculations were limited to those scenarios where low temperatures, low burnups, or both were present.

The bubble concentration equations incorporated the production and loss mechanisms of diffusion, production, resolution, homogeneous nucleation, random gas atom capture, biased gas atom capture (due to thermal gradients), random coalescence, and biased coalescence. In addition, effects due to biased diffusion out of the grain were included in a separate routine which calculated the biased release by the overlap of two spheres shifted by an amount consistent with the bubble velocity and the time period.

The bulk vacancy and interstitial concentration equations incorporated the production and loss mechanisms of diffusion, production, thermal emission from bubbles and dislocations, capture by bubbles and dislocations, and recombination.

The bubble radii equations were derived from considerations of the number of vacancies and interstitials entering and leaving a bubble. Governing processes from the bubble concentration equations were incorporated and modified by the associated vacancies involved during the interactions. In addition, direct interaction with vacancies and interstitials was incorporated through appropriate terms for thermal emission of vacancies and capture of vacancies and interstitials. Originally, every radius dependent term in these equations was updated during iterations. However, instabilities were encountered which were lessened when the terms coming directly from the concentration equation were held constant during iterations. Holding these terms constant did not eliminate all the causes of instability, and further restrictions had to be imposed on bubble radii calculations. The first of these was the enforcement of a minimum radius a bubble could have. This was necessary because of the occasional calculation of negative pressures based on the radius. The minimum radius was determined based on the yield pressure of uranium nitride. The yield pressure was determined from a crude least squares fit to available stress-strain curves. A second restriction imposed was the enforcement of a maximum radius the bubble could have. The maximum radius was the equilibrium radius of the bubble. This avoided a stability problem in the thermal emission term, presumably caused by a loss of too much non-linearity in formulating the correction part of this term. These two additional restrictions stabilized the radii calculations, but added additional constraints on what REDSTONE could model. Steady-state or ramp-up scenarios would still be able to have their overpressurized non-equilibrium bubbles modeled. However, ramp-down scenarios would have their underpressurized bubbles modeled with equilibrium bubbles.

The validity of REDSTONE calculations were checked against analytical solutions for internal consistency and against experimental studies for agreement with the swelling and release results. Internal consistency checks validated that REDSTONE's basic structure was correctly programmed. These checks also indicated that the best agreement with transient results would be obtained by using an element scheme with smaller and smaller elements as the grain surface is approached, and by shifting the middle node in the outermost element towards the grain surface. The checks performed against experimental studies showed REDSTONE was satisfactorily accounting for the intragranular fission gas portion of swelling, but that other non-gaseous sources of swelling accounted for up to 1.5% additional swelling.

Once validated, REDSTONE was used to calculate the swelling and gas release associated with a hypothetical SDI scenario. Since an SDI nuclear system has not been specified, the reactor parameters used here were obtained from the SP-100 nuclear reactor. This reactor has a design failure point of 1% diametral increase of the fuel pin. Operating conditions for the scenario were calculated by a SP-100 systems code developed at Texas A&M University. The SDI scenario involved 10 years of low power operation at 10 kW followed by a 30 second rise to 1 MW. The 1 MW power was maintained for approximately 15 minutes. Fuel temperature was 500°K at low power and 1500°K at high power. The fission gas swelling and release results calculated by REDSTONE indicated that swelling would only be slightly above 1% and release would be negligible. Even when swelling due to non-gaseous components is added to this, the diametral increase in the fuel pellet remains below 1%, with almost 6% needed for design failure.

This same scenario was run at higher temperatures to see if uncertainties in the operating parameters could cause significantly different results. Both the low power and high power temperatures were increased by 100°K, 200°K, and 300°K. The increased temperatures had no significant effect during the low power period. After the high power period, the swelling and release results changed slightly, but the differences were unimportant. However, further investigation into the calculated solutions revealed that the results at the highest temperature of 1800°K appeared unreliable because high bubble mobilities were causing a shift to bubbles larger than those modeled by REDSTOINE. Thus the swelling calculated by REDSTONE was smaller than if all possible bubble groups had been considered. Whether the actual swelling would have been important for failure considerations is not known since REDSTONE is not capable of modeling the bubble sizes required. REDSTONE's

results are important indications, however, since centerline fuel temperatures were expected to be near 1800°K.

It was apparent from REDSTONE's results that further investigations into the swelling behavior near 1800°K were needed. Additional uncertainties in property values used in REDSTONE and in the operating parameters of the fuel might even lower this transition temperature to 1700°K or lower.

Many of the property values used in REDSTONE had to be gleaned from a small library of studies. In some cases, only one value may have been available for use. In other cases, the values had to be taken from other fuels such as UC or UO_2 . As a result, most of the properties have a great deal of uncertainty associated with them. In an effort to understand the effect these uncertainties might have on the solution, several of the properties were varied and the results compared. These included the xenon, surface, and irradiation enhanced diffusion coefficients, the surface energy, and the resolution parameter. Except for surface energy, the properties were varied by two orders of magnitude on either side of the programmed value. Most of the changes caused by varying the properties were small and unimportant with regards to failure. However, each property showed definite trends which might become important at higher temperatures. Thus, knowing these trends is an important design consideration.

The largest effects were observed from changes in irradiation enhanced diffusion and resolution. Although the changes caused by varying these properties were small, both had solutions which exhibited significant shifts to larger bubble sizes. The irradiation enhanced diffusion coefficient produced such results above 100 times the programmed value. The resolution parameter produced such results below 0.01 of the programmed value. However, the resolution parameter used in

REDSTONE is already lower than that experimentally observed for UO_2 , so it is unlikely that the value used in REDSTONE is two orders of magnitude too high. On the contrary, the uncertainties in the irradiation enhanced diffusion coefficient would be more likely to range towards larger values in the same direction as the fall in reliability.

In situations where the bubble sizes remain small, the largest effect of uncertainties in property values could come from the resolution parameter. The value used in REDSTONE is one to two orders of magnitude lower than in UO_2 because it was assumed that UN resolution would occur via a single knock-on mechanism versus a thermal spike mechanism in UO_2 . If, in fact, resolution is the same in both fuels, then the parameter study here indicates a reduction in the swelling and a corresponding increase in release.

Both the irradiation enhanced diffusion and resolution property studies suggested that the bubble profile after the power rise was determined by a combination of the pre-rise profile and the mobility of the bubbles during the high temperature period. Even though the time at high temperatures was short, it appeared that the conditions after the power rise were less a function of the pre-rise conditions than hypothesized.

9.2 Need for Further Work

Throughout this study, various areas, both external and internal to this study, were found which deserve further investigation. External to the study, it was clear that information on microscopic UN properties is severely lacking and much of it dates to the early 1970's and 1960's. Most of the information available is scattered in the literature and not very well corroborated. This includes data on diffusion of

gaseous species, the number of vacancies associated with gaseous species, the effect of stoichiometery, the effect of fission damage, surface energies, and nuclear processes such as resolution mechanisms, fission damage production rate, and nucleation mechanisms. These are clear candidates for additional work. In addition, microscopic studies of swelling in UN needs to be performed so that an intragranular code can be benchmarked directly against the data it produces.

Internal to this study, additional work was identified both by additional questions which arose and by deficiencies which were inevitably incorporated. The additional questions were 'what if' questions dealing with the scenario modeled. The study here looked at a single hypothetical scenario and a few of the parameters which could affect it. There are of course other possible scenarios, all which could be investigated. These other scenarios might involve different power histories as well as other reactor types. Some of the questions cannot be answered until firmer design specifications for an SDI system are developed. Additionally, sensitivity studies such as the ones performed on property uncertainties remain areas for additional work. Besides additional property data, calculational factors such as time step size and element spacing could be investigated. These latter studies are useful for fully characterizing the capabilities of the program.

The deficiencies needing additional work deal with limitations on the type of scenarios which could be modeled. The main limitation was that REDSTONE needed to model each individual bubble size. This prevents the calculation of results in high swelling situations. As a result, first on the list of additional work needed on REDSTONE is the incorporation of a bubble grouping scheme which can model the larger bubbles. The next most important area of improvement in REDSTONE is the calculation of bubble radii changes. As discussed previously, constraints built into

REDSTONE limit is usefulness in modeling underpressurized bubbles. These constraints need to be removed. In order to accomplish this the formulation of the bubble radii equations needs to be reworked to eliminate instabilities. Part of this would have to involve a restatement of the equation of state for xenon so that negative pressures could not be calculated. Possible approaches to rework the radii equations could be as simple as replacing the full equations with a relationship which does not incorporate the terms of the concentration change equation and does not require iteration, or as detailed as finding a way to incorporate all the types of equations into one calculation, and keeping the non-linearities. Many investigators have taken the former approach, and this is probably the easiest to accomplish.

9.3 Conclusions

Several conclusions can be drawn from this study. These are:

- Non-linear finite element techniques can be used successfully to model the transient and non-equilibrium equations describing fission gas behavior in nuclear fuel. The most difficult part of the finite element formulations as performed here was in dealing with the non-equilibrium bubble behavior.
- There is a lack of adequate experimental studies on microscopic swelling of UN to adequately benchmark an intragranular fission gas routine.
- 3) Within the benchmarking limitations, the program developed for this study appears to satisfactorily account for the fission gas component of swelling in the grain. Non-gaseous component swelling of fuel could add up to 1.5% additional swelling at low burnups.

- 4) Many of the property values needed to model fission gas behavior in UN either are not known and must be adopted from other materials or have large uncertainties because of the lack of corroborating studies. Uncertainties in the irradiation enhanced diffusion coefficient and the resolution parameter appear to have the greatest potential for causing significant swelling or release changes at higher burnups and/or higher temperatures.
- 5) The SDI scenario modeled here is not predicted to produce significant fuel swelling or release provided the temperature remains below 1700°K. Between 1700°K and 1800°K, bubble mobilities increase significantly so that accelerated swelling is possible. Uncertainties present in property values and operating parameters could lower this temperature even more.

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APPENDIX 1

PROGRAM REDSTONE LISTING

	ers ers binc, iunits, lout, conlim)	4 ·	÷	*	bed dof	+	+			IF (IREST.EQ.1) CALL RESINP(DENOM1, DNOM2V, DNOM2I, DENOM3)		(VLINC,0)	*	•
CONRAT = 1.00+06 VACRAT = 1.00+06 RADRAT = 1.00+06	<pre></pre>	Generate connectivity arrays	CALL ICOND	Generate J and L arrays	Used for determining prescribed dof	. Find max bandwidth	CALL SEMBND	DENOM1 = 1.000 DNOM2V = 1.000 DNOM2I = 1.000	DENOM3 = 1.000 Atmrel = 0.00	IF (IREST.EQ.1) CALL RESINF	DO 101 IPRD=1,NPRD	DO 102 LINC=1,NLINC(IFRD)	CALL PROP(IPRD,LINC,NLINC,0)	Start iterations	DO 50 ITERG=1,MAXITR ITRFLG = 0
	TTTTTTT 000000 NN NN EEEEEEE TTTTTTT 0000000 NNN NN EEEEEEE TT 00 00 NNN NN EEE	NNN NN 00 00	NN	and stresses using	tment		+ –	- +			, ACCESS=' SEQUENTIAL ' , FORM=' UNFORMATTED')	+			- +
A1.1 Main Program	RR EEEEEEE DUODDDD SSSSSS TITT RR EEEEEEE DUODDDD SSSSSSS TITT RR EEE DD 00 SSSSSSS TITT RR EEE DD 00 SS TIRR EEEE DD 00 SSSSSSS T	5555555 00 00 55 00 00 55 555555 00000000	000000 555555	This program calculates displacements and stresses using finite element algorithms.	Daniel L. Deforest Nuclear Engineering Department Texas A&M University	PROGRAM REDSTONE	Get type declarations, common block,	and initializations INCLUDE (MAIN)	REAL*16 CNVRR(21),CNIRR(21) Logical subinc	CALL XUFLOW(0)	OPEN(UNIT=10,STATUS='NEW',ACCESS='S	Contraction normatant	NLINC - number of load increments LINC - current load increment	CONRAT - initial convergence ratio set high initially CONLIM - limit for determining	convergence

	* Start iterations
alculat: elemental stiffness matricis t and force vectors and assemble into	DO 20 ITERAT = 1,MAXITR
global matricis. The convergence rotio is calculated here based on R-F 1/1 R)	 Calculate elemental stiffness matricis and force vectors and assemble into alabal matricis
CALL ASSEMB(IPRD,LINC,NLINC,ITERAT)	 The convergence ratio is calculated here based on IIR-Fi1//IIRII
ine to solve equations	CALL ASSMBVCIPRD,LINC,NLINC
CALL SULVE (MAXAT, INVARIO) IF (IPRD.NE.1.OR.LINC.NE.1.OR.ITERAT.NE.1.OR.ITERG.NE.1) CONRAT = RNOR4(RRBC, INONPD)/DEMOMI	 Call routine to solve equations
IF (IREST.EQ.1) CONRAT = RNORM(RRBC,INONPD)/DENOM1 Call EXPAND(1)	CALL SOLVE (MAXHV,INONPV) CALL EXPAND(2)
est for convergence. If convergence then Increment lood vector Output results If LINC-MLINC then end	 Test for convergence. If convergence then Increment load vector Output results If LINC=NLINC then end
DENOMI = RNORM(CONC, INONPD)+RNORM(DELCON,INONPD) IF (CONRAT.LT.CONLIM) GOTO 151 ITRFLG = 1	8
Go to next iteration	103 CONIINUE IF (IPRD.NE.1.OR.LINC.NE.1.OR. - VACRAT = RNORM/CNVRR.NN-1)
CONTINUE	IF (IREST EQ.1)
f get to this point then too many iterations	+ VALINA = NUORM(CUTR, NN-1) + RNORM(CUTR, NN-1)
CALL ERROR('Maximum iterations encountered.')	<pre>* Test for convergence. * Tf convergence then</pre>
	 Increment load vector

CALL ASSMBV(IPRD, LINC, NLINC, ITERAT) IF (IREST.EQ.1) VACRAT = RNORM(CNVRR,NN-1)/DNOMZV + RNORM(CNIRR,NN-1)/DNOMZI to solve equations + +----CALL SOLVE (MAXHV, INONPV) CALL EXPAND(2) INC then end nvergence. ence then load vector ults

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<pre>* Output results * If IINC=NIINC then end </pre>	* IF (RADRAT.LT.CONLIM) GOTO 123 ITRFLG = 1	60 to hext iteration	30 CONTINUE	<pre>* If get to this point then too * many iterations</pre>	CALL ERROR('Maximum iterations encountered.')	123 CONTINUE	CALL RADII IF (ITRFLG.EQ.0) GOTO 200 5.000 TATINUE			CALL SETCON(IPRD,LINC,NLINC,1) CALL SETCON(IPRD,LINC,NLINC,2) CALL SETCON(IPRD,LINC,NLINC,2)	CALL OF CONTAINED THE METHOD THE DETEMINATION ATTADED A	CALL BRELL(ATNU) LINC, MLINC, TCINLL, TCI BRE, ATMALL)	IF (((LINC/LOUT(IPRD))*LOUT(IPRD)).EQ.LINC) + CALL OUTPUT(LINC,IPRD,NLINC,.TRUE.,ITERAT,IUNITS,	+ PCTREL, PCTSML)	102 CONTINUE	CALL RESSAVLUENOML, UNUMER, UNUMEL, UENOMS)	9999 END	وتوقيها والمحكم والاماد المحادر الأثاب المحاطة المحادية والمحاد المحادية والمحادية والمحاد والمحاد والمحادة وللمحادة وللمحادة والمحادة و	EEEEEE MMMMM FFFFFFF MMMMMM	SS SS EE
• If LINC-NLINC then end	DNOMZY = RNORW(CONCY,NN-1)+RNORW(DELCNV,NN-1) DNOMZI = RNORW(CONCI,NN-1)+RNORW(DELCNI,NN-1) IF (VACRAT.LT.CONLIM) GOTO 122	Go to next iteration	20 CONTINUE	<pre>* If get to this point then too * many iterations</pre>	Call ERROR('Maximum iterations encountered.')	122 CONTINUE	<pre>\$ \$ tart iterations </pre>	DO 30 ITERAT = 1, MAXITR	CALL ASSMBRCIPRD.LINC.NLINC.ITERAT)	CALL SOLVE (MAXHR, INONPR)	CALL EXPAND(3)	* Test for convergence.	* If convergence then * Increment load vector	* Output results * If INC_NINC then end		e	IF (IKESI.EQ.I) KADKAI = KNOKM(KK8C,INONPK)/DENOM3	UENUM3 = KNUKN(KAU,INUNPK)+KNUKN(UELKAU,INUNPK)	Test for convergence. T f rouvergence.	Increment load vector

<pre>* Get element stiffness matrices and force vectors and create the * global matrix and vector. * The result is stored in banded storage with prescribed * distributed and storage with for</pre>	 alsplucement rows eliminated and accounced for. ELIKF - subroutine that provides elemental values T TAND - row on alchal matrix 	 IDAND - row or growt matrix IBAND - column of global matrix (banded) ELK(i,j) - element K value for ith element row and jth element colul ELF(i) - element F value for ith row UP(i) - prescribed displacement of ith global node 	DO 110 I=1,NE NEDOF = NNELEM*NBBLSZ	3001 CALL ELKFG(I,IPRD,LINC,NLINC,ITERAT) 3999 DO 111 M=1,NEDOF	+ IGME = LUNUP-(I,M) LIGME = LARRAY(IGME) IBAND = LIGME JMAX = LIGME+MAXH-1	JMLN = LIGME-MAXH+1 DO 112 N=1,NEDOF IGNE = ICONDF(I,N) LIGNE = LARRAY(IGNE)	 Do not add in stiffness matrix if either value matches that Of a prescribed dofthese values are subtracted from the force vector later 	IF (LIGNE.EQ.0.OR.LIGME.EQ.0) GOTO 40 JBAND = LIGNE-LIGME+MAXH	<pre>* Don't worry about the lower half of the matrix.its symmetric i *</pre>	IF (LIGNE.LT.JMIN.OR.LIGNE.GT.JMAX) GOTO 40 RKBC(IBAND,JBAND) = RKBC(IBAND,JBAND) + ELK(M,N)
888888 888888 88 8888 88 88 88 8888888 888888		Wop	NUCL(1) - number of degrees of freedom for element i ICONDF(i,j) - connectivity array of local dofs to global dofs i.e. global node number of local node J for element i	ب. ۵		+ - +	f stiffness	+		
W W W W W W W W W W W W W W W W W W W 	bandstorage,	uts are: NE - total number of elements NDOF - total number of degrees of freedom NPD - total number of prescribed degrees of freedom	NUDFL(1) - number of degrees of freedom for element 1 ICONDF(i,j) - connectivity array of local dofs to glo i.e. global node number of local node J for element i	uurruu.) - guodau degree or rreedom number or uu prescribed displacement UP(i) - prescribed displacement of global node i	ERAT)		ero K and R matrices RKBC(i,j) - global value of ith row and jth column of stiffness RRBC(i) - global value of ith row of force vector			
EEEEE EEEEE EEEEEE EEEEEEE	global, trix.	s es of fr ibed deg	or rree rray of de numbe	acement ment of	INC, ITEF		row and w of for	1 7 7 8 8		
\$\$\$\$\$\$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$	mbles the system ma	uts are: NE - total number of elements NDOF - total number of degrees of NPD - total number of prescribed of	ber or aegrees connectivity a i.e. global no for element i	uurruu, - guodal aggree of freedom prescribed displacement UP(i) - prescribed displacement of	RD,LINC,NL	2 2 7 7 7 8 1 8 1 8 1 8 1 8 7	ue of ith of ith ro	1 0 0 0 0 0	908	
\$\$\$\$\$\$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$\$ \$	tine asse condensed	l number tal numbe al number	- number j) - conn i.e. for (global prescri prescribe	ASSEMB(IP	us (NI	atrices lobal value bal value	108 I=1, INONPD D0 109 J=1, 2*MAXH	108 I=1,IDWRB1 D0 109 J=1,IDWRB2 RKBC(I,J) = 0.000	NUE = 0.000
AA AAAAAAAA AAAAAAAA AA AA AA AA AA AA	This subroutine assembles the global, symmetric, condensed system matrix.	Inputs are: NE - tota NDOF - tot NPD - tot	ICONDF(1,	UP(i) - 1	SUBROUTINE ASSEMB(IPRD,LINC,NLINC,IT	Get declarations INCLUDE (MAIN)	Zero K and R matrices Zero K and R matrices RKBC(i,j) - global v RRBC(i) - global val	00 108 I=1,INONP0 D0 109 109 J=1,2*M	DO 108 I-1,IDMRB1 DO 109 J-1,IDM RKBC(I,J) -	CONTINUE RRBC(I) = 0.000 CONTINUE
				• • • •	ļ	Get .	Zer KK		00	109 108

CONTINUE 40

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IF (LIGME.EQ.0) GOTO 111 RRBC(LIGME) = RRBC(LIGME)+ELF(M) CONTINUE CONTINUE

111 110

return End

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SS SS MW MM MW 88 88 SSSSSSS SW MM MW 888888 SSSSSSS SSSSSS MW MM WW 8888888 SSSSSSS SSSSSS MW MM WW 8888888 SSSSSS SSSSS MM MM 88888888 SSSSSS SSSSS MM MM 88888888 SSSSSS SSSSSS MM MM 88888888 Utine assembles the global, bandstorage, condensed system matrix.	AAAAAAA	\$\$\$\$\$\$\$\$	55555555	MMM	MMM		RRRRR	RR	
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This subroutine assembles the global, bandstorage, symmetric, condensed system matrix. Inputs are: NE - total number of elements		\$\$\$\$\$\$	\$\$\$\$\$\$	Ŧ	W	_	RR	RR	
Inputs are: NE - total number of elements	This subro symmetric,	utine asse condensed	mbles the system ma	globa trix.	r, bo	indstorage,			
NE - total number of elements	Inputs are:								
	NE - tot	al number	of element	s.					

NDOF - total number of degrees of freedom NPD - total number of prescribed degrees of freedom NDOFE(i) - number of prescribed degrees of freedom for element i ICONDF(i,j) - connectivity array of local dofs to global dofs i.e. global node number of local node J for element i IDOFPD(i) - global degree of freedom number of ith prescribed displacement

DO 110 I=1,NE NEDOF = NNELEM*NBBLSZ

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Get declarations	•
INCLUDE (MAIN)	+
Zero K and R matrices	+
RKBC(i,j) - global value of ith row and jth column of stiffness RRBC(i) - global value of ith row of force vector	
INONPR = (NN-1)*(NBBLSZ-1) MAXHR = NNELEM*(NBBLSZ-1)	•
DO 108 I=1, INONPR DO 109 J=1, Z*MAXHR RKBC(I, J) = 0.000 109 CONTINUE RRBC(I) = 0.000 108 CONTINUE	
Get element stiffness matrices and force vectors and create the	
grout matrix and vector. The result is stored in banded storage with prescribed displacement rows eliminated and accounted for.	
NEU	,
ELK(i,j) - element K value for ith element row and jth element colu ELF(i) - element F value for ith row UP(i) - prescribed displacement of ith alobal node	t colu

CALL ELKFR(I, IPRD, LINC, NLINC, ITERAT) 3001

DO 111 M-1.NEDOF **666**

LIGNE = LARRAYCIGNE) IF (N-CN/NBBLSZ*NBBLSZ).NE.1.AND.LIGNE.NE.0) THEN LIGNE = (I-1)*CNNELEM-1)*CNBBLSZ-1)-CN-1)/NBBLSZ+N-1 IF (M-(M/NBBLSZ*NBBLSZ).NE.1.AND.LIGME.NE.0) THEN
LIGME = (I-1)*(NNELEM-1)*(NBBLSZ-1)-(M-1)/NBBLSZ+M-1 IGNE - ICONDF(I,N) JMIN = LIGME-MAXHR+1 DO 112 N=1, NEDOF LIGHE - LARRAY(IGHE) JMAX = LIGHE+MAXHR-1 IGME = ICONDF(I,M) LIGNE = 0 ENDIF **IBAND = LIGME** LIGME = 0ELSE ENDIF ELSE

+----Do not add in stiffness matrix if either value matches that of a prescribed dof...these values are subtracted from the force vector later

IF (LIGNE.EQ.0.OR.LIGME.EQ.0) GOTO 40 JBAND = LIGNE-LIGME+MAXHR

 Don't worry about the matrix outside band -----

1 +---

IF (LIGNE.LT. JMIN.OR.LIGNE.GT. JMAX) GOTO 40

RKBC(IBAND, JBAND) = RKBC(IBAND, JBAND) + ELK(M,N)

CONTINUE \$ Subtract prescribed values from force vector if row is not a prescribed dof but column is • . •

+

RKBC(i,j) - global value of ith row and jth column of stiffness RRBC(i) - global value of ith row of force vector 33333 ≥ non-symmetric, condensed system matrix for calculating the ž ≥ 1 33333³ * Add in element force value if not for a prescribed dof vacancy and interstitial concentration in the grain This subroutine assembles the global, bandstorage, 88 88 88888888 888888888 88888888 8888888 8888888 8888888 88 88 SUBROUTINE ASSMBV(IPRD, LINC, NLINC, ITERAT) MANANA RRBC(LIGNE) = RRBC(LIGNE)+ELF(M) MANA IF (LIGME.EQ.0) GOTO 111 \$\$\$\$\$\$\$\$ \$\$\$\$\$\$\$ S \$\$\$\$\$\$\$\$\$ \$\$\$\$\$\$ 5555555 555555 \$\$\$\$\$\$\$\$ \$\$\$\$\$\$\$ \$\$\$\$\$\$\$ S 555555 5555555 555555 Zero K and R matrices CONTINUE CONTINUE INCLUDE (MAIN) Get declarations CONTINUE Inputs are: \$ AAAAAAA ₹ AAAAA RE TURN END ≸ ≸ * 112 111 .

\$\$\$\$\$\$\$ \$\$\$\$\$\$ +----Don't worry about the lower half of the matrix..its symmetric \$\$\$\$\$\$\$ \$\$\$\$\$\$ 5555555 Subtract prescribed values from force vector if row is not a 555555 RKBC(IBAND, JBAND) = RKBC(IBAND, JBAND) + ELK(M,N) + + IF (LIGNE.LT.JMIN.OR.LIGNE.GT.JMAX) GOTO 40 Add in element force value if not for a prescribed dof 3 3 ₿ 3 3 3 UUUUUUU nnnnn This subroutine calculates diffusion coefficients 6 IF (LIGNE.EQ.0.0R.LIGME.EQ.0) GOTO JBAND = LIGNE-LIGME+MAXHV 333333 FFFFFFFF FFFFFFF RRBC(LIGME) = RRBC(LIGME)+ELF(M) FFFF FFFF Ľ 比 出出 IF (LIGME.EQ.0) GOTO 111 FFFFFFF prescribed dof but column is FFFF FFFF Ľ H ĽĽ IIIIIII IIIIIIII IIIIIII IIIIIII • CONTINUE CONFINUE CONTINUE CONTINUE 00000000 0000000 0000000 RETURN END 112 111 ļ \$ IBAND - row or global matrix JBAND - column of global matrix (banded) JELK(i,j) - element K value for ith element row and jth element colul ELF(i) - element F value for ith row Get element stiffness matrices and force vectors and create the of a prescribed dof...these values are subtracted from the Do not add in stiffness matrix if either value matches that The result is stored in banded storage with prescribed 6) | UP(i) - prescribed displacement of ith global node 6 1 ELIKF - subroutine that provides elemental values IF (I.EQ.ME.AND.N.EQ.MEDOF) LIGNE = 0 IF (I.EQ.ME.AND.N.EQ.(NEDOF-1)) LIGNE displacement rows eliminated and accounted for. LIGME = (I-1)*(NEDOF-2)+M IF (I.EQ.NE.AND.M.EQ.NEDOF) LIGME = 0 IF (I.EQ.NE.AND.M.EQ.(NEDOF-1)) LIGME CALL ELKFV(I, IPRD, LINC, NLINC, ITERAT) LIGNE = (I-1)*(NEDOF-2)+N JMAX = LIGME+MAXHV-1 JMIN - LIGME-MAXHV+1 DO 112 N-1,NEDOF RKBC(I,J) = 0.000 global matrix and vector. 00 109 J=1, 2*MAXHV IBAND - LIGME NEDOF = NNELEM^{*}2 DO 111 M-1, NEDOF RRBC(I) = 0.000INONPV = (NN-1)*2 D0 108 I=1,INONPV MAXHV = NNELEM*2 force vector later CONTINUE DO 110 I-1.NE CONTINUE 3999 3 108 •

ļ	SUBROUTINE DIFFUS	EEEEEEEE LLLLLL KK KK FF GGGGGG I	
		 This routine calculates the elemental contributions to fission gas bubbles using spherical geometry. 	
	INCLUDE (LATTICE)	SUBROUTINE ELKFG(IELEM, IPRD, LINC, MLINC, ITERAT)	
	<pre>DIFFX = DIFFXA*QEXPC-DIFFXB/TEMP(1)) + DIFFIE*FISSRC(1) DIFFS = DIFFSA*QEXPC-DIFFSB/TEMP(1)) DIFFU = DIFFUA*PRSNGN(1)**DIFFUB*QEXPC-DIFFUC/TEMP(1)) DIFFN = DIFFNA*PRSNGN(1)**DIFFNB*QEXPC-DIFFNC/TEMP(1)) + GRNDIA**(OIFND1+DIFND2*TEMP(1))*1.Q0 DIFF1 = DIFFIA*QEXPC-DIFFIB/TEMP(1))</pre>	INCLUDE (MAIN) INCLUDE (ELKF) INCLUDE (GAUSS) INCLUDE (PI) INCLUDE (BOLTZMAN) INCLUDE (SURFENGY)	
	PREFIX = 3.Q0%LAWBDA*OMEGA/(2.Q0*PI) Prefix = Prefix*4.Q0/(DiffLM**2)	 Zero element force and stiffness matricis 	
	DO 101 I-1,NN	D0 450 I=1, IDMEK1	
	DIFFCF(I,1) = DIFFX	ELF(L) = 0.000 D0 451 J=1 (DMEK2 ELF(L) = 0.000	
	IF (NBBLSZ.GE.2) THEN DO 102 3-2,NBBLSZ	451 CONTINUE V. 020 450 CONTINUE	
	<pre>DIFFCF(I,J) = DIFFS/(RADIUS(I,J)**2) * PREFIX IF (RADIUS(I,J).LT.(DIFFLM/PI)) DIFFCF(I,J) =</pre>	<pre>* Get global node numbers *</pre>	
102	E CONTINUE ENDIF	NODE1 = ICON(IELEM,1) NODE2 = ICON(IELEM,2) NODE3 = ICON(IELEM,2)	
101	L CONFL/!!)E RETURN END		
ļ	EEEEEEEE LL KK FFFFFFF GGGGGGG EEEEEEEE LL KK KK FFFFFFF GGGGGGGG	DO 701 ITR=1, IGP CALL SHAPE(NNELEM, ITR, IGP, SHPFCT, DERSHP)	
* * *	EE LL XXXXX FFFF EE LL XXXXX FFFFF EE LL XXXXX	 Value of R in real coordinates of integration point 1 	
• •	EE EEEEE IIIIIII KK KK FF GGGGGGG EEEEEEE IIIIIII KK KK FF		

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		CONCIP(N) = SHPFCT(1)*(CONC(IDOF1) + DELCON(IDOF1)) + + SHPFCT(2)*(CONC(IDOF2) + DELCON(IDOF2)) + + DELCIP(N) = SHPFCT(3)*(CONC(IDOF3) + DELCON(IDOF3)) DELCIP(N) = SHPFCT(1)*DELCON(IDOF1) +) +) + DELCON(IDOF1))	+ DERSHP(2)*(CONC(IDDF2) + DELCON(IDDF3)) + DELCON(IDDF3)) +		IF (RAULP(N).61.15.4-8) 1HEN RESOL(N) = RESOLP*0FLOAT(N)* + (1.Q0-(1.Q0-15.Q-8/RADIP(N)-RESADJ)**3)	ELSE RESOL(N) = RESOLP*QFLOAT(N) ENDIF	SRCFRC(N) = 8.40128Q0*PI*OMEGA*RADIP(N)**2/LTPARM**2 TOTSRC = TOTSRC + SRCFRC(N)*CONCIP(N)	THEN	<pre>MKLIE(8,*) Bubble concentrations have exceeded',</pre>	 Start assembly of equations 	DO 101 N=1,NBBLSZ
+ SHPFCT(2)*R(MODE2)+ + SHPFCT(3)*R(MODE3)	 Calculate Jacobian matrix and determinant 1 	AJMAT = DERSHP(1) * R(NODE1) + + DERSHP(2) * R(NODE2) + + DERSHP(3) * R(NODE3)	AJDET = AJMAT	<pre>* Calculate numerical integration multipliers times R**2 </pre>	WMULT = WEIGHT(ITR,IGP)*RGP**2*AJDET	 Calculate source 1 	<pre>*support = ATMSRC(IMAT(IELEM))</pre>	<pre>temp temp temp temp temp temp temp temp</pre>	TOTSRC = 0.00 DO 106 N=1,NBBLSZ	 Calculate integration point bubble radii, diffusion coefficient, 1 concentrations, delta concentrations, derivative concentrations, 1 velocities, resolution parameters 	RADIP(N) = SHPFCT(1)*RADIUS(MODE1,N) + + SHPFCT(2)*RADIUS(NODE2,N) + + SHPFCT(3)*RADIUS(NODE3,N)	<pre>DFCFIP(N) = SHPFCT(1)*DIFFCF(NODE1,N) + + SHPFCT(2)*DIFFCF(NODE2,N) +</pre>

IF (K.LE.NCHECK) ROE = -4.Q0*PI*(RADIP(N)-RADIP(K))
(DFCFIP(N)+DFCFIP(K)) -PI(KADIP(N)+RADIP(K))**2 *QABS(VELIP(N)-VELIP(K)) +----+ ETA = SOURCE*SRCFRC(K) IF (N.EQ.1) THEN IF (K.EQ.2) THEN ETA = 2.00*RESOL(K) ETA = ETA - SOURCE*SRCFRC(K) * Resolution gain and source gain (loss for N=1) ETA = ETA - SOURCE+SRCFRC(K) ENDIF * Coalescence loss, both random and biased j Add in contributions from ETA and ROE IF (K.EQ.(N+1)) THEN ETA = RESOL(K) ETA - RESOL(K) IF (K.EQ.N) GOTO 102 IF (K.EQ.(N-1)) EPSILN = 0.00ROE = 0.00 ETA = 0.00 ELSE -----ENDIF ENDIF ELSE +----******************** * Zero constants • ******** ****** -----+ + + + TELK(N,N) = -1.Q@/DELTIM(IPRD) + BETA + 2.Q@*GAMMA*CONCIP(N)
TELF(N) = DELCIP(N)/DELTIM(IPRD) - ALPHA - BETA*CONCIP(N)
- GAMA*CONCIP(N)*CONCIP(N) IF ((2*N).LE.NBBLSZ) GAMMA = -32.Q0*PI*RADIP(N)*DFCFIP(N) +----+-------- Self-coalescence loss (twice regular coalescence loss) + * Set up re-solution loss and atomic source terms | * Loop over contributions from other bubble groups * Already taken care of process involving only N * Add in terms to matrix along with time terms BETA = -RESOL(N) NCHECK - NBBLSZ - N ALPHA = SOURCE ENDIF +-----DO 102 K=1,NBBLSZ (N.NE.1) THEN BETA = 0.00 * Max bubble size for ROE AL PHA = 0.00 BETA = 0.00 Gamma = 0.00 * Zero constants | +-----ELSE H • *----* ******* ÷

+	TELK(N,N) = TELK(N,N) + ROE*CONCIP(K) TELK(N,K) = ETA + ROE*CONCIP(N) TELF(N) = TELF(N) - ETA*CONCIP(K) - ROE*CONCIP(N)*CONCIP(K)	* * m
•	4	
 Gain fro 	Gain from coalescence of other than N bubbles, random and biased	Add results for this integration point into overall element matrix !
+	J = N-K IF (J.GT.0.AND.K.GE.J) THEN EPSILN = 4.Q0*PI*(RADIP(J) + RADIP(K)) *(OFCFIP(J) + OFCFIP(K))	CALL MTXADD(TELK, ELK, NEDOF, NEDOF, IELEM, + IDMCM1, IDMCM2, IDMCM3, IDMEK1, IDMEK2, IDMEK3, + .FALSE., FALSE.)
+ +	+P1*(RADIP(J)+RADIP(X)**2 *QABS(VELIP(J)-VELIP(K))	 Add load vector into overall force vector
* Add in c	Add in contributions	CALL MTXADD(TELF,ELF,NEDOF,1,IELEM, + IDMCV1,IDMCV2,IDMCV3,IDMEF1,IDMEF2,IDMEF3,
r 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	TELK(N.K) - TELK(N.K) + EPSILN*CONCIP(J)	+
		<pre>* Next integration point *</pre>
102	CONTINUE	701 CONTINUE
<pre> Fill out </pre>	Fill out rest of matrix (other nodes) and include diffusive terms	ENO • • • • • • • • • • • • • • • • • • •
	DO 103 ND1-NNELEM,1,-1	1:
	IND1 = (ND1-1)*NBBL52 + N DO 104 L=1,NBBL52	LL KK KK FF
	DO 105 NDZ=NNELEM,1,-1 INDZ = (NDZ-1)*NBBLSZ + L	EE LL KKKKK FFFF KKKRR EE LL KKKK FFFF RRRRR
+	IF (N.NE.L) THEN TELK(IND1,IND2) = TELK(N,L)*WAULT*SHPFCT(ND1) *SHPFCT(ND2)	EEEEEEEE LLLLILLIK KK FF RR RR EEEEEEEE LLLLILLIK KK FF RR RR
	ELSE TELK(IND1,IND2) = (-DFCFIP(N)/AJDET+*2 *DERSHP(ND1)*DERSHP(ND2)	 This routine calculates the elemental contributions to fission gas bubbles using spherical geometry.
+ +	+TELK(N,L)*SHPFCT(ND1) *SHPFCT(ND2))*MAULT	SUBROUTINE ELKFR(IELEM, IPRD, LINC, NLINC, ITERAT)
105	E MULF CONTINUE	INCLUDE (MAIN)

INCLUDE (ELKF) INCLUDE (GAUSS) INCLUDE (PI) INCLUDE (BOLTZMAN) INCLUDE (SURFEMGY) INCLUDE (VANDMAAL) INCLUDE (LATTICE) INCLUDE (DIFFCOEF) INCLUDE (DIFFCOEF) Zero element force and stiffness matrices	D0 450 I=1,IDMEK1 ELF(I) = 0.000 D0 451 J=1,IDMEK2 ELK(I,J) = 0.000 CONTINUE CONTINUE	DO 453 I=1,IDMCM1 TELF(I) = 0.000 DO 454 J=1,IDMCM2 TELK(I,J) = 0.000 CONTINUE CONTINUE	et global mode numbers { NODE1 = ICON(TELEM,1) NODE2 = ICON(TELEM,2) NODE3 = ICON(TELEM,3)	1218 8	value of R in real coordinates of integration point
 	1 451 450	454 453	et .	<u> </u>	* <a< td=""></a<>

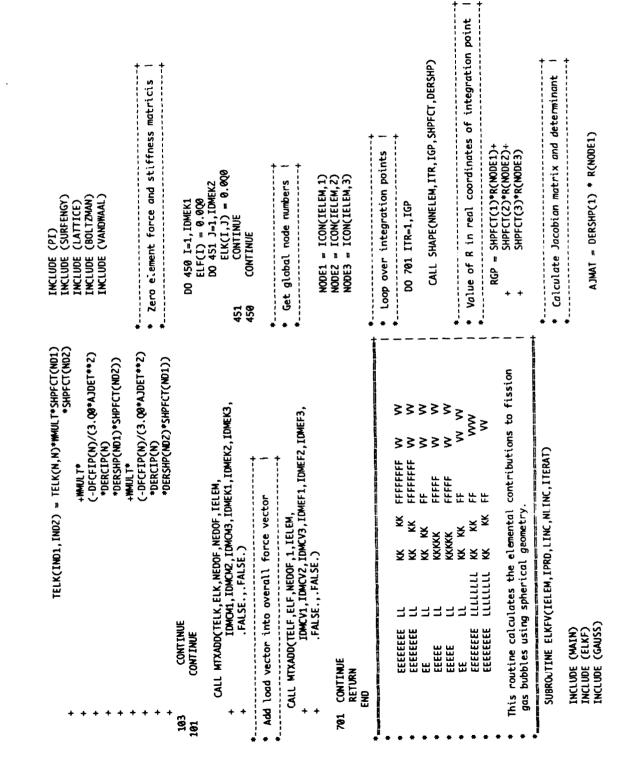
ALPHA = ALPHA +32.00*PI*CONCIP(N)**2*DFCFIP(N)/3.00 BETA = BETA - 32.00*PI*CONCIP(N)**2*DFCFIP(N)/3.00 ALPHA = ALPHA+SRCFRC(N)*CONCIP(N)*CNRDIP(N)/3.00 BETA - 2, Q0*OMEGA*DIFFV*CONCIP(N)*CONVIP *RADIP(N)/CNRDIP(N)**3 ALPHA = ALPHA+RESOL(N)*CONCIP(N)*CNRDIP(N)/3.00 ALPHA = ALPHA-OMEGA*DIFFV*CONCIP(N)*CONVIP BETA = BETA-SRCFRC(N)*CONCIP(N)/3.00 *RADIP(N)*CNRDIP(N) BETA = BETA-RESOL(N)*CONCIP(N)/3.00 *RADIP(N)/CNRDIP(N)**2 *RADIP(N) IF ((2*N).LE.NBBLSZ) THEN TOTSRC = QMINICTOTSRC, 1.Q0) * Set up self-coalescence loss DO 101 N=2, NBBLSZ Set up resolution loss 0000 +--------------ALPHA = 0.Set up source loss CONTINUE ENDIF +-----BETA GAMMA Zero constants BETA Vacancy gain ****** • + + + + 106 . • IF (BPTMIP(N).LT.0.00.0R.BPTMIP(N).GT.YPRS) BPTMIP(N)=YPRS SRCFRC(N) = 8.4012800*PI*OMEGA*RADIP(N)**2/LTPARM**2 IF (BPRSIP(N).LT.0.OR.BPRSIP(N).GT.YPRS) BPRSIP(N)=YPRS ID0F3 · ICONDF(IELEM,(2*NBBLS2+N))
CONCIP(N) = SHPFCT(1)*(CONC(ID0F1) + DELCON(ID0F1)) +
SHPFCT(2)*(CONC(ID0F2) + DELCON(ID0F2)) +
SHPFCT(3)*(CONC(ID0F3) + DELCON(ID0F3))
DELCIP(N) = SHPFCT(1)*DELCON(ID0F1) +
SHPFCT(2)*DELCON(ID0F1) + DERSHP(1)*(CONC(ID0F1) + DELCON(ID0F1)) + DERSHP(2)*(CONC(ID0F2) + DELCON(ID0F2)) + DERSHP(3)*(CONC(ID0F3) + DELCON(ID0F3)) (1.00-(1.00-15.0-8/RADIP(N)-RESADJ)**3) SHPFCT(2)*(RAD(ID0F2) + DELRAD(ID0F2)) + DERSHP(1)*(RAD(ID0F1) + DELRAD(ID0F1)) + DERSHP(2)*(RAD(ID0F2) + DELRAD(ID0F2)) + = 3.Q0 / (BLTZMN*(TEMP(1)**2)*(LTPARM**3)) = SHPFCT(1)*(RAD(ID0F1) + DELRAD(ID0F1)) SHPFCT(3)*(RAD(ID0F3) + DELRAD(ID0F3)) = SHPFCT(1)*DELRAD(ID0F1) + DERSHP(3)*(RAD(ID0F3) + DELRAD(ID0F3)) BPRSA(N) = QFLOAT(N)*BLTZMN*TEMP(1)*3.Q0/(4.Q0*PI) *DIFFS*TMPGR(1)*PI*(RADIP(N)**3)*SRFACT BPRSB(N) = 0FLOAT(N)*VANDWL(2)*3.00/(4.00*PI) BPTMIP(N) = BPRSA(N)/(CNRDIP(N)**3-BPRSB(N)) IOTSRC = TOTSRC + SRCFRC(N)*CONCIP(N) BPRSIP(N) = SHPFCT(1)*BBLPRS(NODE1,N) + SHPFCT(2)*BBLPRS(NODE2,N) + SHPFCT(3)*BBLPRS(NODE3,N) + SHPFCT(2)*RADIUS(NODE2,N) + SHPFCT(2)*DELRAD(ID0F2) + SHPECT(2)*DIFFCF(NODE2,N) SHPFCT(3)*DIFFCF(NODE3,N) DFCFIP(N) = SHPFCT(1)*DIFFCF(NODE1,N) = SHPFCT(1)*RADIUS(NODE1,N) SHPFCT(3)*RADIUS(NODE3,N) SHPFCT(3)*DELCON(IDOF3) SHPFCT(3)*DELRAD(IDOF3) (RADIP(N).GT.15.Q-8) THEN RESOL(N) = RESOLP*QFLOAT(N)* RESOL(N) = RESOLP*QFLOAT(N) Ņ H DRRDIP(N) CNRDIP(N) DERCIP(N) **DLRDIP(N)** RADIP(N) (N) dI I J ENDIF ELSE ч +

BETAK = BETAK + SRCFRC(K)*CONCIP(K)/3.Q0
*CNRDIP(K)**3/CNRDIP(N)**2 ALPHA = ALPHA - SRCFRC(K)*CONCIP(K)/3.Q0 BETAN = BETAN - SRCFRC(K)*CONCIP(K)/3.Q0 *CNRDIP(K)**3/CNRDIP(N)**2 BETAN = BETAN - RESOL(K)*CONCIP(K)/3.Q0 *CNRDIP(K)**3/CNRDIP(N)**2 *2.Q0/CNRDIP(N) ALPHA = ALPHA - RESOL(K)*CONCIP(K)/3.00 = BETAK + RESOL(K)*CONCIP(K)/3.Q0 *CNRDIP(K)**3/CNRDIP(N)**2 + *CNRDIP(K)**3/CNRDIP(N)**2 *CNRDIP(K)**3/CNRDIP(N)**2 *2.Q0/CNRDIP(N) *3.00/CNRDIP(K) *3.Q0/CNRDIP(K) IF (K.EQ.(N-1)) THEN (K. EQ. (N+1)) THEN DO 102 K=1,NBBLSZ AL PHA = 0.00 Betan = 0.00 Betak = 0.00 Betaj = 0.00 EPSILN = 0.00ETA = 0.00BETAK ENDIF ENDIF + Resolution gain Ľ +---- Zero constants Source gain ----------------******** + + + + + + + ----*** • *C2.00*SRFENG/CNRDIP(N)-BPTMIP(N)+PRESS(1))) *C2.00/CNRDIP(N)-2.00*OMEGA*SRFENG/(BLTZMN*TEMP(1) *CNRDIP(N)+*Z)-3.00*OMEGA*BPRSA(N)*CNRDIP(N)+*Z /(BLTZMN*TEMP(1)*(CNRDIP(N)**3-BPRSB(N))**Z)) BETA = BETA-(CONCIP(N)+DELCIP(N)/3.Q0)/DELTIM(IPRD) *RADIP(N)/CHRDIP(N)**2 *QEXP(OMEGA/(BLTZMN*TEMP(1))* (2.Q0*SRFENG/RADIP(N)-BPRSIP(N)+PRESS(1))) AL PHA+OMEGA*DIFFV*CONCIP(N)*CONCVE(1,2) BETA+OMEGA*DIFFV*CONCIP(N)*CONCVE(1,2) ALPHA = ALPHA+OMEGA*CONCIP(N)*DIFFI*CONIIP *RADIP(N)/CNRDIP(N)**2 BETA = BETA+OMEGA*CONCIP(N)*DIFFI*CONIIP *RADIP(N)/CNRDIP(N)**3 * Loop over contributions from other bubble groups *QEXP(ONEGA/(BLTZMN*TEMP(1)) ALPHA = ALPHA+(CONCIP(N)*DLRDIP(N) +CNRDIP(N)*DELCIP(N)/3.00) *RADIP(N)/CNRDIP(N)**2 /DELTIM(IPRD) **Z** TELK(N,N) = BETA TELF(N) = ALPHA NCHECK = NBBLSZ +----* Index limit for ROE ------* Interstitial loss * Time differential 1 ALPHA 1 BETA Vacancy loss -----•••••• ----• + + + + +

TE' "(IND1,IND2) = TELK(N,L)*#MULT*SHPFCT(ND1) *SHPFCT(ND2) +PI*CONCIP(J)*CONCIP(K)/(3. Q0*CNRDIP(N)**2) +PI*CONCIP(J)*CONCIP(K)/(3.00*CNRDIP(N)**Z) *(RADIP(K)+RADIP(J))**Z BETAK+4.Q0*PI*CONCIP(K)*CONCIP(J) BETAJ = BETAJ+4.Q0*PI*CONCIP(K)*CONCIP(J) /(3.Q0*CNRDIP(N)**2) *3.Q0*CNRDIP(J)**3/CNRDIP(J) *3.00*CNRDIP(K)**3/CNRDIP(K) Fill out rest of matrix and include diffusive terms *(DFCFIP(J)+DFCFIP(K)) *(CNRDIP(K)**3+CNRDIP(J)**3) *(DFCFIP(J)+DFCFIP(K)) /(3,00*CNRDIP(N)**2) *(RADIP(J)+RADIP(K)) *(RADIP(J)+RADIP(K)) *QABS(VELIP(K)-VELIP(J)) *QABS(VELIP(K)-VELIP(J)) *OABS(VELIP(K)-VELIP(J)) *(RaDIP(K)+RADIP(J))**2 *(RADIP(K)+RADIP(J))**2 INDZ = (NDZ-1)*NBBLSZ + L TELK(N, J) = TELK(N, J) + BETAJTELK(N,N) = TELK(N,N) + BETANTELK(N,K) = TELK(N,K) + BETAK *3.Q0/CNRDIP(K) *3.00/CNRDIP(J) DO 103 ND1=NNELEM,1,-1 IND1 = (ND1-1)*NBBLSZ + N *2.00/CNRDIP(N) TELF(N) = TELF(N) + ALPHADO 105 ND2=NNELEM,1,-1 IF (N.NE.L) THEN *CNRUIP(K)**3 *CNRDIP(J)**3 DO 104 L=1, NBBLSZ ELSE BETAK CONTINUE ENDIF + + + + + + *----102 *(CNRDIP(K)**3+CNRDIP(J)**3)/CNRDIP(N) -PI*CONCIP(J)*CONCIP(K)/(3.Q0*CNRDIP(N)**2) -PI*CONCIP(J)*CONCIP(K)/(3.00*CNRDIP(N)**2) ALPHA = ALPHA+4. Q0*PI*CONCIP(N)*CONCIP(K)/3. Q0 BETAN = BETAN-4.00*PI*CONCIP(N)*CONCIP(K)/3.00 +PI*CONCIP(N)*CONCIP(K)/3.Q0 *(RADIP(N)+RADIP(K))**2 +PI*CONCIP(N)*CONCIP(K)/3.00 ALPHA = ALPHA-4.00*PI*CONCIP(K)*CONCIP(J) /(3.00*CNRDIP(N)**2) *(CNRDIP(K)**3+CNRDIP(J)**3) BETAN = BETAN-8.00*PI*CONCIP(K)*CONCIP(J) *QABS(VELIP(N)-VELIP(K)) *QABS(VELIP(N)-VELIP(K)) *(RADIP(N)+RADIP(K))**2 *(DFCFIP(J)+DFCFIP(K)) *(DFCFIP(N)+DFCFIP(K)) *(DFCFIP(N)+DFCFIP(K)) *(DFCFIP(J)+DFCFIP(K)) *(CNRDIP(K)**3+CNRDIP(J)**3) /(3.00*CNRDIP(N)**2) *(RADIP(J)+RADIP(K)) *(RADIP(J)+RADIP(K)) *(RADIP(N)+RADIP(K)) (KADIP(N)+RADIP(K)) *QABS(VELIP(K)-VELIP(J)) *(RADIP(K)+RADIP(J))**2 IF (J.GT.0.AND.K.GE.J) THEN *CNRDIP(N) *CNRDIP(N) IF (K.LE.NCHECK) THEN [F (K.EQ.N) GOTO 102 ENDIF +----+------7-2 = C +-------+----Coalescence gain | Coalescence loss |

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SHPFCT(2)*B8LPRS(NODE2,N) + SHPFCT(3)*B8LPRS(NODE3,N) IF (BPRSIP(N).LT.@.OR.BPRSIP(N).GT.YPRS) BPRSIP(N)=YPRS CONCIP(N) = SHPFCT(1)*(CONC(ID0F1) + DELCON(ID0F1))
SHPFCT(2)*(CONC(ID0F2) + DELCON(ID0F2))
SHPFCT(3)*(CONC(ID0F3) + DELCON(ID0F3)) alphav = Alphav + 4.00*PI*RADIP(N)*CONCIP(N)*DIFFv ALPHAV = SOURCE + ZVAC*DIFFV*CONCVE(1,2)
*QEXP(-OMEGA*PRESS(1)/(BLTZMN*TEMP(1))) Set up source and equilibrium gain from dislocations 1 +----+----+ + BPRSIP(N) = SHPFCT(1)*BBLPRS(NODE1,N) RADIP(N) = SHPFCT(1)*RADIUS(NODE1,N) SHPFCT(2)*RADIUS(NODE2,N) SHPFCT(3)*RADIUS(NODE3,N) *QEXP(OMEGA/(BLTZMN*TEMP(1)) IDOF1 = ICONDF(IELEM,N) IDOF2 = ICONDF(IELEM,(NBBLSZ+N)) IDOF3 = ICONDF(IELEM,(Z*NBBLSZ+N)) Add in equilibrium gain from bubbles = - ZVAC*DIFFV *CONCVE(1,2) = - ZINT*DIFFI ALPHAI = SOURCE + +----------ALPHAV = 0.00 ALPHAI = 0.00 BETAV = 0.00 BETAI = 0.00 DO 106 N=2, NBBLSZ * Loop over matrix rows BETAV BETAI *----* *----* 4 + + + + + + + + + + ÷ + + + +

DERSHP(1)*(CONCT(NODE1) + DELCNT(NODE1)) +
DERSHP(2)*(CONCT(NODE2) + DELCNT(NODE2)) +
DERSHP(3)*(CONCT(NODE3) + DELCNT(NODE3)) SHPFCT(2)*(CONCI(NODE2) + DELCNI(NODE2)) +
SHPFCT(3)*(CONCI(NODE3) + DELCNI(NODE3)) CONVIP = SHPFCT(1)*(CONCV(NODE1) + DELCNV(NODE1)) SHPFCT(2)*(CONCV(NODE2) + DELCNV(NODE2)) SHPFCT(3)*(CONCV(NODE3) + DELCNV(NODE3)) SHPFCT(1)*(CONCI(NODE1) + DELCNI(NODE1)) DERSHP(1)*(CONCV(NODE1) + DELCNV(NODE1)) DERSHP(2)*(CONCV(NODE2) + DELCNV(NODE2))
DERSHP(3)*(CONCV(NODE3) + DELCNV(NODE3)) Calculate numerical integration multipliers times R**2 SHPFCT(2)*DELCNV(NODE2) + SHPFCT(3)*DELCNV(NODE3) SHPFCT(1)*DELCNI(NODE1) + SHPFCT(1)*DELCNV(NODE1) + SHPFCT(2)*DELCNI(NODE2) SHPFCT(3)*DELCNI(NODE3) MMULT = WEIGHT(ITR,IGP)*RGP**2*AJDE1 + DERSHP(2) * R(NODE2) + DERSHP(3) * R(NODE3) SOURCE - VACSRC(1) Calculate source = AJMAT H H 1 . 8 CONITP DELVIP DERVIP DERIIP DELIIP AJOET + + + + + + + + + + + #

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*********** *----*

Zero constants

+ +

IND1A = (ND1-1)*2+1 IND1B = (ND1-1)*2+2 DO 105 ND2-HNELEM,1,-1 IND2A = (ND2-1)*2+1 IND2A = (ND2-1)*2+2 IND2B = (ND2-1)*2+2 TELK(IND1A, IND2A) = (-DIFFV/AJDET**2 *5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(1, 1)*5HPFCT(ND1) +TELK(2, 2)*5HPFCT(ND1) +TELK(2, 2)*5HPFCT(ND1) +TELK(2, 2)*5HPFCT(ND2) +TELK(2, 2)*5HPFCT(ND2)	<pre>TELK(IND1A,IND28) = TELK(1,2)*HMULT*SHPFCT(ND1) TELK(IND18,IND2A) = TELK(2,1)*HMULT*SHPFCT(ND2) TELK(IND1A) = (DIFFV*DERSHP(ND1)/AJDET*2 TELF(IND1A) = (DIFFV*DERSHP(ND1)/AJDET*2 TELF(IND1B) = (DIFFI*DERSHP(ND1)/AJDET*2 *DERIIP+SHPFCT(ND1)*TELF(1))*HMULT TELF(IND1B) = (DIFFI*DERSHP(ND1)/AJDET*2 *DERIIP+SHPFCT(ND1)*TELF(2))*HMULT CONTINUE CONTINUE CALL MTXADDCTELK,ELK,NEDOF,NEDOF,IELEM, IDMCM1,IDMCM2,IDMCM3,IDMEK1,IDMEK2,IDMEK3, .FALSE.,FALSE.) </pre>	Add load vector into overall force vector CALL MTXADD(TELF,ELF,NEDOF,1,IELEM, + IDMCV1,IDMCV2,IDMCV3,IDMEF1,IDMEF2,IDMEF3, + .FALSE.,.FALSE.) 701 CONTINUE RETURN END	EEEEEEE QQQQQQ UU UU RRRRRR AAAAA DDDDDDD EEEEEEE QQQQQQ UU UU RRRRRRR AAAAAA DDDDDDDD EE QQQQQQ UU UU RR RR AA AA DD DDDDDDD EEEE QQ UU UU RR RR AA AA DD DD
<pre>+ *(2.Q0*SRFENG/RADIP(N)-BPRSIP(N)+PRE5S(1))) Include loss to bubbles and dislocations + BETAV = BETAV - + 8ETAI = BETAV - + 4.Q0*PI*RADIP(N)*CONCIP(N)*DIFFV + 4.Q0*PI*RADIP(N)*CONCIP(N)*DIFFI + 1 4.Q0*PI*RADIP(N)*CONCIP(N)*DIFFI + 4.Q0*PI*RADIP(N)*CONCIP(N)*CONCIP(N)*DIFFI + 4.Q0*PI*RADIP(N)*CONCIP(N)*DIFFI + 4.Q0*PI*RADIP(N)*CONCIP(N)*CO</pre>	<pre>Start assembly of elemental stiffness matrix and force vector 1 TELK(1,1) = -1.00/DELTIM(IPRD) + BETAV TELK(2,2) = -1.00/DELTIM(IPRD) + BETAV TELF(2) = DELVIP/DELTIM(IPRD) - ALPHAI - BETAV*CONVIP TELF(2) = DELVIP/DELTIM(IPRD) - ALPHAV - ALPH</pre>	Add in contributions from ROE * Add TELK(1,1) = TELK(1,1) + ROE*CONLIP TELK(2,2) = TELK(2,2) + ROE*CONVIP TELK(1,2) = ROE*CONVIP TELK(2,1) = ROE*CONVIP TELF(1) = TELF(1) - ROE*CONVIP*CONLIP TELF(2) = TELF(2) - ROE*CONVIP*CONLIP	Fill out rest of matrix and include diffusive terms { DO 103 ND1=NNELEM,1,-1

CALL ERROR('Maximum Iterations in EQURAD') 102 EQURAD = RNEW RETURN END	EEEEEEEE RRRRRRR RRRRRRR 000000 RRRRRR EEEEEEE RRRRRRR RRRRRRR 00000000 RKRRRRR EE RR RR 00 00 RR RR EEEEE RR RR RR 00 00 RR RR EEEEE RR RR RR 00 00 RR RR	EE RR RR RR RR 000000 EEE RR RR RR 00000000 EEE RR RR RR 00000000 EEE RR RR RR 0000000000	SUBROUTINE ERROR(A)	CHARACTER*40 A WRITE(8,999) 999 FORMAT(/' ************************************	WRITE(8,998) A 998 FORWAT('',A50) WRITE(8,997) 907 FORMAT('************************************		EEEEE XX XX PPPPPPP AAAAAA NN NN EEEEEE XX XX PPPPPPPP AAAAAAA NNN NN	PPPPPPP AA AA MINNIN PPPPPPP AAAAAAA NN NN PPPPPPP AAAAAAAA NN NN K PP AA AA AA NN
EFEEE QQ QQ UU UU RRRRRRR AAAAAAA DD DD E Ee QQ QQQQ UU UU RRRRRR AAAAAAA DD DD 1 Eefeeeee QQQQQQ UUUUUU RR RR AA AA DDDDDDDD 1 Eefeeeee QQQQQQ UUUUUU RR RR AA AA DDDDDDDD 1	Calculates equilibrium bubble sizes Input is N - the number of atoms in bubble TEMP - temperature of the bubble PRESS - external pressure ROLD - previous radius	REAL FUNCTION EQURAD(N,TEMP,PRESS,ROLD) IMPLICIT REAL*16 (A-H,O-Z) IMPLICIT INTEGER (I-M)	INCLUDE (PI) INCLUDE (BOLTZMAN) INCLUDE (SURFEMGY) INCLUDE (VANDMAAL)	ITMAX = 1000 CONLIM = 1.Q-10 AI DHA = 4 00*DT*DDESS/3 00	ALFINA = 4.00°TITTRESJOJJOU BETA = 8.00°SRFENG4PL/3.00 GAMA = -QFLOAT(N)*(PRESS*VANDML(Z) + BLTZMN*TEMP) ROE = -Z.Q0*SRFENG*VANDML(Z)*QFLOAT(N)	DO 101 I=1,ITMAX RNEW = ROLD - (ALPHA*ROLD**4 + BETA*ROLD**3 + GAMMA*ROLD + ROE) + / (4.Q0*ALPHA*ROLD**3 + 3.Q0*BETA*ROLD**2 + GAMMA)	RATIO = QABS((RNEW-ROLD)/RNEW) IF (RATIO.LT.CONLIM) GOTO 102	ROLD = RNEW 101 CONTINUE

(11111 WRITE(8,1) æ KR KR . R ENDIF FFFFFFF FFFFFFF RETURN END • • . FFFF FFFF Ш *** + 105 104 *** Takes the solved nonprescribed displacements and adds them into IJ = (I-1)*NBBLSZ + J
IF ((RAD(I))+DELRAD(I))+RRBC(INC)).LT.RADMIN(J))
RRBC(INC) = RADMIN(J) - RAD(IJ) - DELRAD(IJ)
IF ((RAD(IJ)+DELRAD(IJ)+RRBC(INC)).GT.RADMAX(J)) 0000000 the prescribed displacement vector UP. Keeps track of IF (CONCV(I)+DELCNV(I)+RRBC(INC+1)).LT.0.Q0)
RRBC(INC+1) = -CONCV(I) - DELCNV(I)
IF ((CONCI(I)+DELCNI(I)+RRBC(INC+2)).LT.0.Q0)
RRBC(INC+2) = -CONCI(I) - DELCNI(I)
DELCNV(I) = DELCNV(I) + RRBC(INC+1)
DELCNI(I) = DELCNI(I) + RRBC(INC+2) ž IF ((CONC(IP)+DELCON(IP)+RRBC(I)).L1.0.Q0)
RRBC(I) = -CONC(IP) - DELCON(IP)
DELCON(IP) = DELCON(IP) + RRBC(I) Z 8 AA DO 105 J-2, NBBLSZ delta Q's, and total Q's SUBROUTINE EXPAND(ITYPE) INC = INC + 1 4 (ITYPE.EQ.3) THEN IF (ITYPE.EQ.1) THEN DO 102 I=1,INONPD (ITYPE.EQ.2) THEN DO 103 I=1,NN-1 IP = JARRAY(I) INC = (I-1)*2 DO 104 I=1,NN-1 ž CONTINUE CONTINUE INCLUDE (MAIN) × Get declarations INC = 0 EEEEEE ENDIF ENDIF H H ÷ + 103 • 102 .

Prints the nodal displacements and elemental stresses, strains 99999999 99999999 0000 0000 ខូ RRBC(INC) = RADMAX(J) - RAD(IJ) - DELRAD(IJ) DELRAD(IJ) = DELRAD(IJ) + RRBC(INC) 000000 ខូន ي ខ្ល AAAAAAA AAAAAAAA \$\$ ₹ AAAAAAA ≸ AAAAA \$\$ \$\$ ddddddd dddddd TTTTTT ===== RR RR RR RR RRRRRRR RRRRRRR RRRRRR RR RR RR RR RR RR ж ж RRRRRRR RRRRRR CONTINUE SUBROUTINE FRTPAG CONTINUE

EEEEEE \$\$\$\$\$\$\$\$\$ SS EEEEEEE \$\$\$\$\$\$\$ \$\$\$\$\$\$\$\$\$ \$\$\$\$\$\$ \$\$\$\$\$\$\$ EEEEEE \$\$\$\$\$\$ EEEEEEE EEEEE S Ш Ш 00000000 8 8 00000000 Ž ž ž NN ¥ ¥ 0000000 0000000 NNN N B 8 ž ž ž ž EEEEEE EEEEEE EEEEEEE EEEEEEE EEEEE 30 00 8 8 8 00000000 0000000 EEEEEE 00 00 000000 00000 33 Ш 8 8 8 FORMATC'1'////// , TT / RRRRRRR RR RR ž ž RRRRRRR י ווווווו R RRRRRRR / RRRRR F F F F E

EE		uts it into nd outputs	, LOUT, CONLEM)	+ ·	+	+ -	*	
UU 99999999999999999999999999999999999	NUUUUU UUUUUUU UUU AAAAAAAA NNN NN III UUU AAAAAAAA NNN NN III	 SUBROUTINE INPUT reads in the input data and p arrays, generates nodes and elements, etc., a input values 	* SUBROUTINE INPUT(MAXITR, NLINC, SUBINC, IUNITS	• Get declarations	INCLUDE (MAIN) INCLUDE (INOUT) LOGICAL SUBINC	 Get date and time of run for output 	IDATE(1) = -1 C CALL DATIM(IDATE) IF (IDATE(1).EQ1) THEN DATE = '*** ** **** TIME = '**:** **'	INUEX = ((IUMIE(LZ)-I)'9)+I DATE(15:23) = DAYS(INDEX:INDEX+8) DATE(7:7) = ','
	eedom		- +	4 -			1 + (1-((,	
NN NN 0000000 NN NN 0000000 NN NN 00 000 NN NN 00 000 NN NN 00 00 NN NN 00 NN 00 NN NN 00 NN 00	This subroutine calculates the degree of fre connectivity arrays	SUBROUTINE ICOND	Get declarations		LUDMAR - CONNECTIVITY GOT GIRAY 100FPD - prescribed dof array UP - displacement vector (including prescribed dofs	DO 101 I=1,NE DO 104 J=1,NBBLS2*4	DO 102 J=1, NBBLSZ DO 102 J=1, NBBLSZ DO 103 L=1, NBBLSZ INDEX = NBBLSZ*(J-1)+L ICONDF(I, INDEX) = NBBLSZ*(CONTINUE	101 CONTINUE
	CCCCCCCC 0000000 NNN NN DDDDDDDDDD CC CC 00 00 NNN NN DD DD0 CC CC 00 00 NN NN DD DD0 CC CC 00 00 NN NNN DD DD0 CCCCCCC 0000000 NN NN DDD0DDDD IIIIIIII NNN NN PPPPPPP UU UU CCCCCCC 0000000 NN NN DD0DDDDD IIIIIIII NNN NN PPPPPPP UU UU CCCCCCC 0000000 NN NN DD0DDDDDD IIIIIIII NNN NN PPPPPPP UU UU	NNN NN D0D0D0DD RETURN NNNN ND DO D ENO NN NN D DO U NN NN DO D U U NN NN DO D U U U NN NN D D U U U U NN NN D D U U U U U NN NN NN NN NN NN PPPPPP U U NN NN NN NN NN NN V U U NN NN NN NN NN N V	NN NN D000000 END NN NN D0 00 0 END NN NN D0 00 1 END NN NN D0 00 1 END NN NN D0 00 1 END NN NN D0 00 00 1 END NN NN NN PPPPPP UU UU NN NN PPPPPPP UU UU NN NN PPPPPP UU UU NN NN PPPPPPP UU UU NN NN PPPPPP UU UU NN NN PPPPPPP UU UU NN NN PPPPPPPPPP UU UU NN NN PPPPPPP UU UU NN NN PPPPPPPPPPPPPPPP UU UU NN NN PPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPP	I CCCCCCC 0000000 WIN WN DDDDDDDD CC CC 00 00 NN NN DD DD DD CC CC 00 00 NN NN DD DD DD CC CC 00 00 NN NN DD DD I CCCCCC 000000 NN NN DDDDDDDD I CCCCCC 000000 NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD I CCCCCC 000000 NN NN DDDDDDDD CCCCCC 000000 NN NN DDDDDDDD CCCCCC 000000 NN NN DDDDDDDD ND CCCCCC 000000 NN NN DDDDDDDD ND ND	I CCCCCCC 000000 NNN NN DDDDDDDD CC CC 00 00 NNN NN DD DD CC CC 00 00 NN NN DD DD CC CC 00 00 NN NNN DD DD CC CC 00 00 NN NNN DD DD CCCCCCC 000000 NN NN DDDDDDD I CCCCCCC 000000 NN NN DDDDDDDD I CCCCCCC 000000 NN NN ND DDDDDDDD I CCCCCCC 000000 NN NN ND DDDDDDDD I CCCCCCC 000000 NN NN DDDDDDDD I CCCCCCCC 000000 NN NN NN DDDDDDDD I CCCCCCC 000000 NN NN NN DDDDDDDDD I CCCCCC 000000 NN NN NN DDDDDDDDD I CCCCCCC 000000 NN NN NN DDDDDDDD I CCCCCCC 000000 NN NN NN DDDDDDD I CCCCCCC 000000 NN NN NN DDDDDDD I CCCCCCCC 000000 NN NN NN DDDDDDD I CCCCCCC 000000 NN NN NN DDDDDDD I CCCCCCC 000000 NN NN NN DDDDDD I CCCCCCC 000000 NN NN NN DDDDD I CCCCCCC 000000 NN NN NN DDDDDD I CCCCCCC 000000 NN NN NN DDDDDD I CCCCCCCCCC 000000 NN NN NN DDDDDD I CCCCCCC 000000 NN NN NN DDDDDD I CCCCCCC 000000 NN NN NN DDDDDD I CCCCCCCC 000000 NN NN NN DDDDDD I CCCCCCCC 000000 NN	I CCCCCCC 000000 NNN NN DD DDDDDDD CC CC 00 00 NNN NN DD DD CC CC 00 00 NN NNN DD DD CC CC 00 00 NN NNN DD DD CC CC 000000 NN NN DD DD I CCCCCCC 000000 NN NN DDDDDDD I CCCCCCC 000000 NN NN DDDDDDD CCCCCCC 000000 NN NN DDDDDDD CCCCCCC 000000 NN NN DDDDDDD I CCCCCCC 000000 NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD NN NN DDDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD NN NN DDDDDDDD CCCCCCC 000000 NN NN NN DDDDDDDD NN NN DDDDDDDD CCCCCCC 000000 NN NN NN DDDDDDDD NN NN DDDDDDDDD CCCCCCC 000000 NN NN NN DDDDDDDD NN NN DDDDDDDDD NN NN DDDDDDDD	I CCCCCCC 000000 NNN NN DD DDDDDDD CC CC 00 00 NN NN DD DD CC CC 00 00 NN NN DD DD CC CC 00 00 NN NN DD DD CC CC 000000 NN NN DD DD CCCCCCC 000000 NN NN DDDDDDD CCCCCCC 000000 NN NN DDDDDDDD CCCCCCC 000000 NN NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD NN NN DDDDDDDD CCCCCCC 000000 NN NN DDDDDDDD NN NN DDDDDDDD NN NN DDDDDDDD	I CCCCCCC 000000 NNN NN DD DD DC CC 00 00 NNN NN DD DD CC CC 00 00 NN NNN DD DD DC CC CC 00 00 NN NNN DD DD DD CC CC CC 00 00 NN NNN DD DD DD DC CC CCCCC 000000 NN NN DD DDDDDD CC CCCCCC 000000 NN NN DD DDDDDD CC CC CCCCC 000000 NN NN DD DDDDDD CC CCCCCC 000000 NN NN DD DDDDDD DD CC CCCCCC 000000 NN NN DD DDDDDD CCCCCCC 000000 NN NN DD DDDDDD CC CCCCCC 000000 NN NN DD DDDDDD CCCCCC 000000 NN NN DD DDDDDD CCCCCCC 000000 NN NN NN DD DDDDD CCCCCCC 000000 NN NN DD DDDDD DD CCCCCCC 000000 NN NN DO DDDDDD CCCCCCC 000000 NN NN NN DD DDDDDD CCCCCCCC 000000 NN NN NN DD DDDDD DD CCCCCCCC 000000 NN NN NN DD

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+ 4-----1 WRITE(UNIT=TIME(5:5),FMT='(11)') IDATE(4) TIME(4:4) = '0' RITE(UNIT=TIME(4:5), FMT='(I2)') IDATE(4) DATE(13:13) = ','
INDEX = ((IDATE(7)-1)*3)+1
INDEX = ((IDATE(7)-1)*3)+1
DATE(1:3) = MONTHS(INDEX:INDEX+2)
WRITE(UNIT=DATE(5:6),FMT='(I2)') IDATE(6)
WRITE(UNIT=DATE(9:12),FMT='(I2)') IDATE(8)
WRITE(UNIT=TIME(1:2),FMT='(I2)') IDATE(10) CALL RDCARD(BUFFER, ICRD, NOFLDS, IFLD, IEND, 0) INDEX = ((IDATE(11)-1)*2)+1 TIME(7:8) = AMPM(INDEX:INDEX+1) ENDIF Only non-comment cards are returned Initialize error checking indicis IF (IDATE(10).GE.10) THEN RDCARD also outputs lines read Read next line of input deck IF (IEND.EQ.1) GOTO 1101 If end of deck goto 1101 TIME(3:3) = ':' -----ENDIF INNAT = 0 INNAT = 0 INPRD = 0 IPRD = 1 ELSE INNFOR = 0 INNTEM = 0 IHDR = 0 INN = 0 INCON = 0 INPRP = 0 INNBOD = 0 INDSP = 0 INNTR = 0 • • • • • • • • • • • ----• • + .

IF (ICRD.NE.@) THEN
CALL ERROR('List of input deck flag not found. ')
ELSE
READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) IOUT
ENDIF

2225 CALL RDCARD(BUFFER, ICRD, NOFLDS, IFLD, IEND, 0)

IF (IEND.EQ.1) GOTO 1101 IF (IEND.EQ.1) GOTO 1101 Goto appropriate input for specific card numbers IF (ICRD/100.NE.1) GOTO 3333 IF (ICRD/100.NE.1) GOTO 3333 ICRD/100.NE.1] GOTO 3235 ICRD/100.NE.1] GOTO 3235 ICRD/100.NE
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WRITS(8,1999) 1999 FORMAT(/' ',80('='))

IF (ICRD.EQ.0) GOTO 1111 GOTO (5001,5002,5003,5004,5005,5006,5007,5008,5009,5010, 5011) ICRD/100 1111 CALL RDCARD(BUFFER, ICRD, NOFLDS, IFLD, IEND, IOUT) CALL ERROR('Unidentified card number in 200s IF (IOUT.EQ.1) WRITE(8,4444) 4444 FORWAT('1'//30X,'LISTING OF INPUT DECK'//) Start output of input deck in form read CALL ERROR('Unidentified card number 5002 G0T0 (5102,5202,5302,5402) ICR0-200 Only non-comment cards are returned Goto appropriate input for specific RDCARD also outputs lines read IGP - Number of Guass points Read next line of input deck * If no math then unknown card IF (IEND.EQ.1) GOTO 1101 If end of deck goto 1101 IUNITS - 1=British 2=Metric Read header info 5001 IHDR = IHDR+1 card numbers G0T0 1111 -----Read in •

+CALL ERROR('Incorrect number of fields on card 300 READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) NE READ(UNIT=BUFFER(FFLD(1)+1:IFLD(2)),FMT=*) NBBL5Z READ(UNIT=BUFFER(IFLD(2)+1:IFLD(2)),FMT=*) NNELEM READ(UNIT=BUFFER(IFLD(2)+1:IFLD(3)),FMT=*) NSPC READ(UNIT=BUFFER(IFLD(3)+1:IFLD(5)),FMT=*) NPRD READ(UNIT=BUFFER(IFLD(5)+1:IFLD(5)),FMT=*) MAXITR NN = NE * (NNELEM - 1) + 1 5102 READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) IUNITS 5302 READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) IDECK READ(UNIT=BUFFER(1:IFLD(1)), FMT=*) IREST ICON(I, J) = (I-1) * (NNELEM-1) + J NBBLSZ - number of bubble groups to track NWAT - number of material types NSPC - number of specified concentrations READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) IGP NNELEM - number of nodes per element Read in connectivity for local nodes NLINC - number of load increments total number of elements NPRD - number of time periods total number of nodes NDOF = NN * NBBLSZ NEDOF = NNELEM * NBBLSZ DO 302 J=1, NNELEM Four numbers required 5003 IF (NOFLDS, NE.6) IMAT(I) = 1CONTINUE DO 301 I-1,NE CONTINUE INPRP = NE GOTO 1111 GOT0 1111 GOTO 1111 GOT0 1111 GOT0 1111 NMAT = 1 Read in: ۳Ň 5202 5402 301

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<u></u> <u>.</u> +---READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) TIMLEN READ(UNIT=BUFFER(IFLD(1)+1:IFLD(2)),FMT=*) NLINC(INPRD) READ(UNIT=BUFFER(IFLD(2)+1:IFLD(3)),FMT=*) LOUT(INPRD) PRESS1(i,j) - initial press for material j and period i POWER1(i,j) - initial power for material j and period i +CALL ERROR('Incorrect number of fields on cards 1000 TEMP1(i,j) - initial temp for material j and period i +CALL ERROR('Incorrect number of fields on cards 800 +CALL ERROR('Incorrect number of fields on cards 900 READ(UNIT-BUFFER(1:IFLD(1)),FMT=*) POROS(INMAT) READČUNIT-BUFFER(JFLD(1)+1.TFLD(2)),FMT+*) ENRICH READCUNIT-BUFFER(JFLD(2)+1.TFLD(3)),FMT+*) CONLIM ENDTIM(INPRD) = ENDTIM(INPRD-1) + TIMLEN POROS(i) - initial porosity for material i Read in for power history for materials DELTIM(i) - time step for period i Read in period characteristics ENDTIM(INPRD) = TIMLEN IMAT(1) = 1 IF (NE.EQ.1) GOTO 77 DO 77 I=2,NE Read in initial porosity IF (INPRD.EQ.1) THEN INMAT = INMAT+1 5010 INPRD = INPRD + 1 5008 IF (NOFLDS.NE.NE) IF (NOFLDS.NE.3) IF (NOFLDS.NE.3) IMAT(I) = 1CONTINUE INPRP = NE GOTO 1111 GOTO 1111 GOT0 1111 ENDIF ELSE 88 27 * . + CALL ERROR('Incorrect number of coordinate fields on cards 500') READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) R(INN) ſ <u>_</u> FORWATC' INDSP = ',I3,' NPD = ',I3) FORMATC' NPD = ',I3,' NPD = ',I3) READCUNIT-BUFFER(IFLDC1)+1:IFLDC2)),FMT=*) IDIR(INDSP) READCUNIT-BUFFER(IFLDC2)+1:IFLDC3)),FMT=*) SPDISP(INDSP) +CALL ERROR('Incorrect number of fields on cards 600 READ(UNIT=BUFFER(1:IFLD(1)),FMT=*) NPD(INDSP) +CALL ERROR('Incorrect number of fields on cards 700 *************************** SPDISP(i) - value of prescribed concentration READ(UNIT=BUFFER(1:IFLD(1)), FMT=*) GRNDIA IDIR(i) - bubble group of prescribed conc. NPD(i) - node of prescribed concentration IMAT(i) - type of material for element i R(I) - R coordinate for global node i Read in prescribed displacement data ICON(i,j) - connectivity for local Values in order of element number node j of element i **GRNDIA** - grain diameter 5007 IF (NOFLDS.NE.1) IF (NOFLDS.NE.1) IF (NOFLDS.NE.3) Read in coordinates 5004 INCON = INCON+1 5006 INDSP = INDSP+1 GOT0 1111 5005 INN = INN+1 G0T0 1111 GOT0 1111 GOT0 1111 Read in Read in 9596 9595 .

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***** 2010 115 . . ? READ(UNIT-BUFFER(IFLD(4)+1:IFLD(5)),FMT=*) POWER2(IPRD,IN+WAT) READ(UNIT-BUFFER(IFLD(5)+1:IFLD(5)),FMT=*) TEMP2(IPRD,IN+WAT) READ(UNIT-BUFFER(IFLD(6)+1:IFLD(7)),FMT=*) TMPGR2(IPRD,IN+WAT) READ(UNIT-BUFFER(IFLD(7)+1:IFLD(8)),FMT=*) PRESS2(IPRD,IN+WAT) READ(UNIT=BUFFER(TFLD(1)+1:IFLD(2)),FMT=*) TEMP1(IPRD,INHMAT) READ(UNIT=BUFFER(IFLD(2)+1:IFLD(3)),FMT=*) TMPGR1(IPRD,INHMAT) READ(UNIT=BUFFER(IFLD(3)+1:IFLD(4)),FMT=*) PRESS1(IPRD,INHMAT) , 15/ <u>5555</u> , IS/ 2 . READ(UNIT=BUFFER(1:IFLD(1)), FMT=*) POWER1(IPRD, INHMAT) 8 . +CALL ERROR('Incorrect number of fields on cards 1100 Total number of specified displacements TEMP2(i,j) - final temp for material j and period i PRESS2(i,j) - final press for material j and period i POWER2(i,j) - final power for material j and period 1101 WRITE(8, 2002) NE, NN, NBBLSZ, NMAT, NSPC, NPRD, MAXITR **fotal number of material types** Maximum bubble groups per node Total number of time periods Maximum number of iterations 4=Nonlinear plane stress Total number of elements - 1-Linear plane stress Total number of nodes IF (NOFLDS.NE.4, AND.NOFLDS.NE.8) NLINC - number of load increments 3=Axisymmetric , UNITS(0, IUNITS) total number of elements - total number of nodes 5011 IF (INHMAT.EQ.MMAT) THEN UNITS(IUNITS) - 1=British 2-Metric IF (NOFLDS.GT.4) THEN =Beam INHMAT = INHMAT+1 IPRO = IPRO + 1INHMAT = 1 2002 FORMAT('1'/' PROBLM(IPRBTP) G0T0 1111 ENDIF ELSE ENDIF Output + ₩₹

<u></u> +CALL ERROR('Incorrect number of coordinate sets entered. - ',1PG10.3E1,' microns'//) 2222 FORMAT(' Number of Guass integration points is ',I2) 2011 FORMAT('1'//26X,'NODAL POSITIONS FROM CENTER'// (', A6, ')'/ : NNETYP(i) - number of nodes for type i element **A8** Output element property data with reference CRDUNT(1, IUNITS, IPRBTP) FORMAT(27X, I5, 6X, 4X, 1PG12.6E1) Units are in ',A7//) WRITE(8,2011) CRDSYS(1,IPRBTP), 26X,'Global Node>I '+',25X,' Check for correct amount of data 2032 FORMAT(//' Grain Diameter IGP - Number of Guass points WRITE(8,2010) I,R(I) Output nodal coordinates Output connectivity data WRITE(8,2032) GRNDIA WRITE(8, 2222) IGP IF (IPR8TP.EQ.3)
. WRITE(8,2222) IGP 26X to material type IF (INN.NE.NN) DO 115 I-1, NN CONTINUE IPRBTP = 1••••••••• Output Set + +

WRTTE(8.2031)	2031 FORMAT('1',29X,'TIME HISTORY CONDITIONS'// +'','Time Initial Initial Initial', + ','Incr Power Temp Press', +'','' Power Temp Press'/ +'+','' Power Temp Press'/	 Check for correct amount of data 	IF (INMAT.NE.NMAT) +CALL ERROR('Incorrect number of material property sets entered. ')	D0 136 I=1,NPRD D0 135 J=1,NMAT MRITE(8,2030) I,POWER1(I,J),TEMP1(I,J),PRESS1(I,J), + POWER2(I,J),TEMP2(I,J),PRESS2(I,J), 2030 FORMAT('',I3,2X,6(1X,1P610.3E1))	136 CONTINUE	 Output material properties 	WRITE(8,2071) 2071 FORMAT('1',24X,'TIME STEP INFORMATION'// +'',16X,'Time Length Number of Time Step Printout'/ +''16X,'Tirr Ascronds) Time Steps (seconds) Therval'/	15X,	 Check for correct amount of data) rect number of material prope	ENDTIM(0) = 0.000 DO 166 I=1,NPRD DELTIM(I) = (ENDTIM(I)-ENDTIM(I-1))/REAL(NLINC(I))	WRITE(8,2230) I,ENDTIM(I),NLINC(I),DELTIM(I),LOUT(I) 2230 FORMATC' ',15X,I3,5X,1PG10.3,2X,I6,6X,1PG10.3,4X,I3)
 NDOF - total number of dofs 	<pre>MRITE(8, 2021) 2021 FORWAT('1'//30X,'ELEMENT CONNECTIVITY'// 2021 FORWAT('1'//30X,'ELEMENT CONNECTIVITY'//</pre>	NDOF = NBBLSZ*NN	 NDOFE(i) - num of dofs for element i 	DO 125 I=1.NE MRITE(8,2020) I,(ICON(I,J),J=1,4) 2020 FORMAT(17X,I5,3X,4(3X,I5)) 125 CONTINUE	 Output element property data with reference to material type 	WRITE(8, 4421)	4421 FORMAI ('1',2/X,'ELEMENI MALEKIAL PRUPENILES'// + 25X,' Material Initial '/ + '+',24X,'	 Check for correct amount of data 	<pre>IF (INPRP.NE.NE) +CALL ERROR('Incorrect number of element property sets entered.')</pre>	DO 425 I=1.NE WRITE(8,5420) I,IMAT(I),POROS(IMAT(I)) 5420 FORMAT(25X,I5,3X,I5,8X,1PG10.4E1)		 Output material properties

ICONDF(i,j) - connectivity array of local dofs to global dofs J array with its values indicating what position a dof occupies among all the NON-PRESCRIBED dofs. Size = total dofs. Zeros Create the L array. This array is basically the inverse of the +-----Here we create the J array which contains indices of global .e. global node number of local node J NPD - number of prescribed degrees of freedom IDOFPD(i) - location in global dof of ith prescribed dof JARRAY(i) - values of J array IDOFPD(i) - global degree of freedom number of ith prescribed displacement of global node i INONPD - total number of non-prescribed dofs degrees of freedom which are not prescribed. as values indicate that dof is prescribed. prescribed displacement ----------------NDOF - total global degrees of freedom ICOUNT = 0
D0 101 I=1,ND0F
D0 102 J=1,NSPC
IF (I.EQ.ID0FPD(J)) G0T0 101 Size = Total dof - Prescribed dof LARRAY(i) - values of the L array for element i JARRAY(ICOUNT) = I ICOUNT = ICOUNT+1 SUBROUTINE INTARS INONPD = ICOUNT CONTINUE INCLUDE (MAIN) **Get declarations** CONTINUE - (i)an 102 101 * . ÷ <u></u> \$\$\$\$\$\$ \$\$\$\$\$\$\$ \$\$\$\$\$\$\$ \$\$\$\$\$\$ \$\$\$\$\$\$ NDOFE(i) - number of degrees of freedom for element i ŝ FCALL ERROR('Incorrect number of displacements entered. NDOF - total number of degrees of freedom NPD - total number of prescribed degrees of freedom ж ж RRRRRRR Rrrrr R 8 Value'/ RRRRRRR RRRRRR This subroutine creates the J array and L array **%** % R Ř WRITE(8,2050) NPD(I), IDIR(I), SPDISP(I) FORMAT(' ',17X, I7,10X, I5,8X,1P610.3) AAAAAAA AAAAAAAA **8** 8 **8** 8 AAAAAAA AAAAA A AA A A NE - total number of elements ппп типт Check for correct amount of data * Output specified displacements. ********* ===== IF (NSPC.NE.0) THEN WRITE(8,2051) IF (INDSP.NE.NSPC) NN NN NNNN DO 155 I=1,NSPC N N ¥ ¥ ¥ CONTINUE CONTINUE Inputs are: IIIIIII IIIIIII IIIIIIII 1111111 H H RETURN Η ENDIF H BD • 2050 155 166

, IBELEM)			UU UU TT UUUUUUU TT UUUUUU TT I stresses, strains	TERAT, IUNITS,	+ +	- +
DIMENSION B(IERDIM, IECDIM, IEDDIM) IAELEM = 1 IBELEM = 1 IF (AFLAG) IAELEM = IELEM IF (BFLAG) IBELEM = IELEM DO 101 I=1,IROM DO 102 J=1,ICOL B(I,J,IBELEM) = A(I,J,IAELEM)+B(I,J,IBELEM)	CONTINUE CONTINUE RETURN END	qqqqqqqq TT UU UU 0000000 qq qq TT UU UU 0000000	00 UU UU TT 00000 UUUUUUU TT 00000 UUUUUUU TT 8 the modal displacements an	SUBROUTINE OUTPUT(LINC, IPRD,MLINC, ITEAD, ITERAT, IUNITS + PCTREL, PCTSML) Get declarations	INCLUDE (MAIN) INCLUDE (INOUT)	<pre>* Calculate percent of load * TIMINC = QFLOAT(LINC)*DELTIM(IPRD) TIMPRD = ENDTIM(IPRD-1) + TIMINC</pre>
DO 103 I-1, NDOF LARRAY(I) = 0 CONTIMUE DO 104 I-1, INONPD IST = JARRAY(I) LARRAY(IST) = I CONTINUE RETURN END	MAMM TTITTTT XX XX MAMMM TTITTTT XX XX MM MM TT XX XX XX XX TT MM MM	00 00 00000 000000 00000 000000 1s them 1s them		IACDIM - the actual column dimension of A in storage I *= IBRDIM - the actual row dimension of B in storage I IBCDIM - the actual column dimension of B in storage AFLAGTRUE. if dimensioned for num of elements I *- BFLAGTRUE. if dimensioned for num of elements I *-	SUBROUTINE MTXADD(A,8, IROW, ICOL, IELEM , IARDIM, IACDIM, IADDIM, IBRDIM, IBCDIM, IBDDIM , AFLAG, BFLAG)	IMPLICIT REAL*16 (A-H,O-Z) IMPLICIT INTEGER (I-N) ŁOGICAL AFLAG, BFLAG

This subroutine sets the properties for all the elements in this 2 WRITE(9,60) I,R(I),J,CONC(IND),RADIUS(I,J),BBLPRS(I,J) (pa)'/ WRITE(8,6) J,CONC(IND),RADIUS(I,J),BBLPRS(I,J) FORMAT(14X,14,3(3X,1P613.6)) ddddddd dd dd (,....., ddddddd dddddd dddddd FORMAT(14,1X,E13.6,1X,14,1X,3(E13.6,1X)) CONTINUE 6 ይ 88888 FORMAT(/13X,'------ Node = ',12,' 'Radius = ',1PG10.3,'----0000000 00000 00000 SUBROUTINE PROP(IPRD,LINC,NLINC,ITER) Ē RR RR RR RR RRRRRR R RRRRRRR R IND = (I-1)*NBBLSZ + 3 RRRRRRR RRRRR RR RR RR R WRITE(8,7) I,R(I) (cm-3) dd ddddddd dd dd dd dd dd dd dd dd (DIFFCOEF) (BOL TZMAN) (SURFENGY) (VANDWAAL) dddddd (LATTICE) DO 2 J=1,NBBLSZ '+',12X, ENCLUDE (MAIN) (Id) DO 1 I=1,NN CONTINUE subperiod INCLUDE INCLUDE INCLUDE INCLUDE 'Group **ENCLUDE** RETURN END 8~ Q 1PG9.3,1X,1PG9.3,1X,1PG9.3,1X,1PG9.3,1X,1PG9.3,1X,0PF9.2) IF (ITEND) WRITE(8,888) IPRD,LINC,DELTIM(IPRD),TIMINC,TIMPRD, IF (ITEND) WRITE(9,6888) IPRD, LINC, DELTIM(IPRD), TIMINC, TIMPRD, Bubble Pressure'/ 16X, Time Step Size = ',1PG10.4E1,' seconds'/
16X, Time in Period = ',1PG10.4E1,' seconds'/
16X,' New Time = ',1PG10.4E1,' seconds',0P//
16X,'Time step represents ',I3,'% of full time period.'// BURATE, BUANTL BUANTL, BUANTL, BUANTL, BUANTL, BUANTL FORWAT(I3,1X,I3,1X,1PG10.4,1X,1PG10.4,1X,1PG10.4,1X,I3,1X/ 16X,'Burnup rate = ',1PG9.3,' atom percent/day'/
16X,'Burnup for period = ',1PG9.3,' atom percent'/
16X,'Burnup (total) = ',1PG9.3,' atom percent',0P/
16X,'Burnup (total) = ',F9.2,' MMD/MT'//) 16X,'Percent fission gas released = ',1PG10.4,'%'/ 16X,'Percent swelling = ',1PG10.4,'%'/// BURATE, BUAPPR, BUAPTL, BUMMTL 16X,'Temperature = ',G10.4,'K')
IF (ITEND) WRITE(9,6887) POWER(1),PRESS(1),TEMP(1) IF (ITEND) WRITE(8,8837) POWER(1),PRESS(1),TEMP(1)
FORMAT(' '//16X,'Power density = ',G10.4,' W/cm3'/
+ 16X,'Pressure = ',G10.4,' Pa'/ IPRCNT, PCTREL, PCTSWL **IPRCNT**, PCTREL, PCTSWL WRITE(8,100) FORMAT(' '//29X,'NODAL CONCENTRATIONS'//) **Bubble Radius** * Output how many iterations and percent load -`. " [PRCNT = 100*LINC/NLINC(IPRD) Concentration 16X, TIME PERIOD ', I3,/ 16X,' Time Step ',I3// * Write out displacements 6887 FORMAT(3(610.4,1X)) 8888 FORMATC'1'// WRITE(8,101) 13X, 101 FORMAT(13X ********* + 'Bubble 4 6888 8887 100

RESADJ = 15.Q-8/VANDWL(2)*(BLTZMN*TEMP(IMTL)/(2.Q0*SRFENG)) TRAD = (3.Q0*OMEGA*VACS/(4.Q0*PI))**0.33333333 ENDIF +-----* Do checking to make sure the above values are what is wanted Set general initial bubble radii and vacancy concentrations! AND. IREST. NE.1) THEN Calculate resolution parameter and resolution adjustment IF (IPRD.EQ.1.AND.LINC.EQ.1.AND.ITER.EQ.0. FISSRC(IMTL) = POWER(IMTL) * 3.1211010 ATMSRC(IMTL) = FISSRC(IMTL) * 0.2500 VACSRC(IMTL) = FISSRC(IMTL) * 5.03 INDEX = (INODE-1)*NBBLSZ+IBBL RESOLP = ((TEMP(IMTL)-500.Q0)/1000.Q0 *(RESLML-RESLMH)+RESLML) rprs = 1.05*(YPSLPE*TEMP(IMTL)+YPINT) VACS = 2*QFLOAT(IBBL) CONC(INDEX) = 0.000 DELCON(INDEX) = 0.000 DELRAD(INDEX) = 0.000 resolp = QMINICRESOLP, RESLMH) RESOLP = RESOLP*FISSRC(IMTL) RESOLP = QMAX1(RESOLP, RESLML) CONCV(NN) = CONCVE(IMTL, Z) CONCI(NN) = CONCVE(IMTL, 1) (IBBL.EQ.1) THEN DO 103 INODE=1,NN DO 102 IBBL=1,NBBLSZ ELSE H for large bubbles ENDIF *********** -----+ + + ZVAC = (2.00*PI*DISDEN)/(QLOG(1.00/(QSQRT(PI*DISDEN)*DISRAD))) LTPARM = LTPRMA + LTPRMB*TEMP(1) + LTPRMC*(TEMP(1)**2) TWPGR(IMTL) = TWPGR1(IPRD,IMTL) +(TWPGR2(IPRD,IMTL)-TWPGR1(IPRD,IMTL)) *QELOAT(LINC)/QFLOAT(NLINC(IPRD)) PRESS(IMTL) = PRESS1(IPRD,IMTL) +(POWER2(IPRD, IMTL)-POWER1(IPRD, IMTL)) +(PRESS2(IPRD, IMTL)-PRESS1(IPRD, IMTL)) +----+----+(TEMP2(IPRD,IMTL)-TEMP1(IPRD,IMTL)) *OFLOATCLINC)/QFLOATCNLINC(IPRD)) POFLOAT(LINC)/OFLOAT(NLINC(IPRD)) QFLOAT(LINC)/QFLOAT(NLINC(IPRD)) Calculate sink strengths of dislocations for vacancies CONCVE(IMTL, 2) = QEXP(-EFORMV/TEMP(1))/OMEGA Calculate scaled power, temp, temp gradient, pressure associated fission gas and vacancy production rates. Calculate equilibrium vacancy and interstitial concs POWER(IMTL) = POWER1(IPRD, IMTL) TEMPICIPRD, IMTL) AWBDA = LTPARM*0.62996052500 OMEGA = (LTPARM**3)*.2500 ********** PRSNGNCIMTL) = 1.00 CONCVE(IMTL,1) = 0.00 IF (ITER.EQ.0) THEN Calculate fission rate and ZINT = 1.0200*ZVAC DO 101 IMTL-1, NMAT **TEMP(INTL)** and interstitials ·····

≂ RADIUS(INODE,IBBL)	+s	+	11111111 11111111 0000000 11111111 11111111 00000000 11111111 11111111 00000000 111 111 11111111 00000000 111 1111111111	TINE RADII E (MAIN) E (PI) E (PI) E (SURFENGY) E (SURFENGY) E (SURFENGY) E (SURFENGY) E (SURTAAN) E (SURTAAN) E (SULTAAN) D=2,NBBLSZ J=2,NBSZ J=2,NBSZ J=2,NBSZ J=2,NBSZ J
RAD(INDEX) = 103 CONTINUE 102 CONTINUE	ENDIF Calculate diffusion coefficients	CALL DIFFUS CALL DIFFUS 101 CONTINUE RETURN END	RRRRRR RRRRRR RRRRRR RR AAAAAAA DDD RR RR AA AAAAAA DD RR RR AA AA DD RRRRRR AAAAAAA DD RRRRR AAAAAAA DD RR RR AAAAAAA DD RR RR AA AAAAAA DD RR RR AA AA DDC RR RR AA AA DDC	SUBROUTINE RADII INCLUDE (MAIN) INCLUDE (PI) INCLUDE (SURFENGY) INCLUDE (SURFENGY) INCLUDE (BOLTZMAN) INCLUDE (BOLTZMAN) INCLUDE (VANDWAAL) DO 10 J=2,NBBLSZ DO 10 J=2,NBBLSZ DO 20 I= 1,NN INDEX = (I-1)*NBBLSZ+J RADIUS(I,J) = 3,Q0*OFL + (4,Q0*DI* + IF (BBLPRS(I,J).1).11.0.Q
Set values for node on grain boundary (equilibrium values) If (INODE.EO.NN.AND.IBBL.NE.1) THEN	RADMAX(IBBL) = EQURAD(IBBL,TEMP(IMTL),PRESS(IMTL),TRAD) RADMIN(IBBL) = (0.75Q0*QFLOAT(IBBL)/PI * CITYMATTHUOC1)/VPRS	<pre>+VANDWL(2)))**0.3333333333333300 * +VANDWL(2)))**0.333333333333300 * RADIUS(INODE,IBBL) = RADMAX(IBBL) BBLPRS(INODE,IBBL) = 3.Q0*QFLOAT(IBBL) *BLTZMN*TEWP(1)/ *BLTZMN*TEWP(1)/ *BLTZMU*TEWP(1)/ *BLPZMU*US(INODE,IBBL)**3 - 3.Q0*QFLOAT(IBBL)*VANDWL(2))</pre>	<pre>IF (BBLPRS(INODE,IBBL).GT.YPRS.OR. + BBLPRS(INODE,IBBL).LE.0.Q0) THEN BBLPRS(INODE,IBBL) = YPRS BBLPRS(INODE,IBBL) = (0.75Q0*QFLOAT(IBBL) + YANDWL(21))**0.3333333333333333333333333333333333</pre>	IF (IBBL.NE.1) THEN BBLPRS(INODE,IBBL) = 3.Q0*GFLOAT(IBBL) *BLTXM*TEMP(1)/ (4.Q0*PTTRAD**3 - 3.Q0*OFLOAT(IBBL)*VANDML(2)) IF (BBLPRS(INODE,IBBL).GT.YPRS.OR. BBLPRS(INODE,IBBL).GT.YPRS.OR. BBLPRS(INODE,IBBL) = YPRS RADIUS(INODE,IBBL) = YPRS RADIUS(INODE,IBBL) = 0(0;7500*GFLOAT(IBBL) /PI*(BIL)=0,0(0;15BL) = YPRS RADIUS(INODE,IBBL) = 0(0;7500*GFLOAT(IBBL) /PI*(BIL)=0,0(0;15BL) = 7RAD ELSE RADIUS(INODE,IBBL) = TRAD ENDIF ELSE RADIUS(INODE,IBBL) = TRAD RADIUS(INODE,IBBL) = TRAD RADIUS(INODE,IBBL) = TRAD RADIUS(INODE,IBBL) = TRAD RADIUS(INODE,IBBL) = TRAD
s for m			ues for a	

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This subroutine saves the restart file.	SUBROUTINE RESSAV(DENOM1, DNOM2V, DNOM2I, DENOM3)	INCLUDE (MAIN)	REWIND(10) WDTTEZION NE NNEIEN NRPLSZ	MRITE(10) TOTATM WRITE(10) DENOMI, DNOMZV, DNOMZI, DENOM3		IND = (I-1)*NBBLSZ WRITE(10) (CONC(J),J=IND+1,IND+NBBLSZ)	WRITE(10) (RAD(3),J=IND+1,IND+NBBLSZ) WRITE(10) (RADIUS(I,J),J=1,NBBLSZ) WDITE(10) (PDI DEC(T 1),J=1 UDBLSZ)	MRITE(10) (WRITE(10) (CONTINUE RETURN END		RRRRR NN NN 000000 RRRRF Rrrrr nnn nn 0000000 rrrrr Rr nnnn nn 00 00 rr	RR NN NN NN OO 00 00 00 00 00 00 00 00 00 00 00 00	00 0000 RR RR MM MM 000000 RR RR MM MM 000000 RR RR MM 000000 RR MM	e norm of a vector.	Input is VECTOR - the array containing the vector LENGTH - length of array (INTEGER)	REAL FUNCTION RNORM(VECTOR,LENGTH) REAL*16 VECTOR(1050) INTEGER LENGTH
* * * *								10	<u>.</u>	• • •	* * •	• • •	••	* * *	
RR RR RRRRRRR RRRRR	¥ %				0PEN(UNIT=11,STATUS='OLD',ACCESS='SEQUENTIAL',FORM='UNFORMATTED')			(' 1117 ' FILE ')					+4 60 44 54 55 55 54 54	N	्र्ट्रेड्र
žźź			ENON3)		IAL', FORM-			NBBLST.ME.NBBLSZ) IT MATCH INPUT FILE		(Z+ON				AAAAA	алицица Ал ал Алалала Алалала Ал Ал Ал Ал Ал Ал Ал Ал Ал Ал Ал Ал
===		art file.	V, DNOMZI, E		S∽' SEQUENT					ND+NBBLSZ) = ', CONC(1 D+NBBLSZ)	188L SZ) 188L SZ)			\$\$\$\$\$\$	ssssss ssssss ssssss ssssss sssssss ssss
\$\$\$\$\$\$\$ \$\$\$\$\$\$ \$\$	\$\$\$\$\$\$\$ \$\$\$\$\$\$	is the rest	NOM1, DNOM2		OLD', ACCES	ET,NBBLST , NET	', NNELET	LET.NE.NNE TART FILE LV,DNOM2I,	2),	(RADIUS(I,J),J=1,NBBLSZ) (BBLPRS(I,J),J=1,NBBLSZ)	CONCV(I), CONCI(I)		\$\$\$\$\$\$	ssssss ssssss ssssss ssssss sssssss ssss
	EEEEEEE	tine input	SUBROUTINE RESINP(DENOM1, DNOM2V, DNOM21, DENOM3)	(NIR	1, STATUS='	READ(11) NET, NNELET, NBBLST PRINT *. ' NET - '.NET	NNELET - NBBLST -	IF (NET.NE.NE.OR.NNELET.NE.NNELEM.OR. CALL ERROR(' RESTART FILE DOES NO READ(11) RESATM READ(11) DENOM1, DNOM2V, DNOM21, DENOM3	10 I = 1,NN IND = (I-1)*NBBLSZ	READ(11) (CONC(J), J=IND+1, IND+NBBLSZ) PRINT *, ' CONC(', IND+2,') = ', CONC(IND+2) READ(11) (RAD(J), J=IND+1, IND+NBBLSZ)) (RADIUS() (BBLPRS(EEEEEE	EEEEEEE EEEEEE EEEEEE EEEEEEEEEEEEEEEE
RR RRRRR RRR	RR RR RR	This subroutine inputs the restart fil	SUBROUTINE	Include (main)	OPEN(UNIT=1	READ(11) N PRINT *.		IF (NET.NE.NE.O + CALL ERROR READ(11) RESATM READ(11) DENOM1	DO 10 I = 1,NN IND = (I-1)	READ(11 PRINT * READ(11	READ(11) READ(11)	KEAD(11) CONTINUE RETURN	END	1	RR RR RR RR RR RR RR RR RR RR RR RR RR R
• • • •	• • •		ļ									10			••••

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SS HH HH AA AA PP PP EE SSSSSS HHHHHHH AA AA PP PP EEEEE SSSSSSS HHHHHHH AA AA PP PPPPPP EEEEE SSSSSSS HHHHHHH AAAAAAA PPPPPPP EEEEE SSSSSSS HH HH AAAAAAA PPPPPPP EE SSSSSSS HH HH AA AA PP EEEEEEE SSSSSSS HH HH AA AA PP EEEEEEE	This routine calculates the shape functions and derivitive shape functions SUBROUTINE SHAPE(NNELEM, ITR, IGP, SHPFCT, DERSHP)	ect - GAL	SHPFCT(2) = 0.000 SHPFCT(2) = 0.000 SHPFCT(3) = 0.000 DERSHP(1) = 0.000 DERSHP(2) = 0.000 DERSHP(3) = 0.000 DERSHP(4) = 0.000	 Calculate value of shape function at coordinates of integration point IF (NNELEM.EQ.2) THEN SHPFCT(1) = 0.5Q0*(1.Q0-RNAT(ITR, IGP)) SHPFCT(2) = 0.5Q0*(1.Q0+RNAT(ITR, IGP)) 	ENDIF IF (NNELEM.EQ.3) THEN SHPFCT(2) = 1.00-RNAT(ITR,IGP)**2 SHPFCT(2) = 0.500*(1.00-RNAT(ITR,IGP)) - 0.500*SHPFCT(2) SHPFCT(3) = 0.500*(1.00-RNAT(ITR,IGP)) - 0.500*SHPFCT(2) ENDIF IF (NNELM.EQ.4) THEN IF (NNELM.EQ.4) THEN	SHPFCI(2) = 1.40-MAN(LIN,144)7-2 SHPFCT(1) = 0.500*(1.00-NNAT(ITR,1GP)) - 0.500*SHPFCT(2) SHPFCT(4) = 0.500*(1.00-NNAT(ITR,1GP)) - 0.500*SHPFCT(2) SHPFCT(3) = -1.687500 * RNAT(ITR,1GP)**3
SSSSSSS EEEEEEE TT CCCCCCC 0000000 NN NN I SSSSSSS EEEEEEE TT CCCCCCC 0000000 NN NN I This subroutine sets the properties for all the elements in this I subperiod. SUBROUTINE SETCON(IPRD,LINC,NLINC,ITYPE)	INCLUDE (MAIN) INCLUDE (LATTICE) INCLUDE (PI)	IF (ITYPE.EQ.1) THEN DO 101 I-1,NN*NBBLSZ CONC(I) = CONC(I) + DELCON(I) DELCON(I) = 0.0Q0 101 CONTINJE	IF (ITYPE.EQ.Z) THEN DO 103 I=1,NN-1 CONCV(I) = CONCV(I) + DELCNV(I) CONCI(I) = CONCI(I) + DELCNI(I) DELCNV(I) = 0.0Q0 DELCNI(I) = 0.0Q0 DEL	IF (ITYPE.EQ.3) THEN DO 104 I=1,NN-1 DO 105 J=2,NBBLSZ IJ = (I-1)*NBBLSZ + J RAD(IJ) = RADIUS(I,J)	DELRAD(IJ) = 0.000 105 CONTINUE 104 CONTINUE 104 ENDIF RETURN RETURN	SSSSSS HH HH AAAAA PPPPPP EEEEEE SSSSSSS HH HH AAAAAA PPPPPPP EEEEEE SSSSSSSS HH HH AAAAAAA PPPPPPP EEEEEEE

(a)(b)(c	<pre>Solution is obtained using Guassian elimination and back substitution. Reference algorithms can be found in THE FINITE ELEMENT METHOD by 0.C. Zienkiewicz 0.C. Zienkiewicz</pre>
+1.687500 • NUAT(ITR, IGP) +0.562500 • NUAT(ITR, IGP) +0.662500 • NUAT(ITR, IGP) +0.62500 • NUAT(ITR, IGP)	(NNELEM. EQ. 2) THEN DERSHP(1) = -0.500 DERSHP(2) = 0.500 ENDIF (NNELEM. EQ. 3) THEN DERSHP(2) = -2.00 * RNAT(ITR, IGP) DERSHP(3) = 0.500 - 0.500*DERSHP(2) DERSHP(3) = 0.500 - 0.500*DERSHP(2) DERSHP(2) = -2.00 * RNAT(ITR, IGP) DERSHP(2) = -2.00 * RNAT(ITR, IGP) DERSHP(2) = -2.00 * RNAT(ITR, IGP) DERSHP(2) = -0.500 - 0.500*DERSHP(2) DERSHP(2) = -0.500 - 0.500*DERSHP(2) DERSHP(2) = -0.500 * 0.500*DERSHP(2) DERSHP(3) = -5.062500 * RNAT(ITR, IGP) + 1.687500 DERSHP(3) = -5.062500 * RNAT(ITR, IGP) + 1.687500 DERSHP(4) = DERSHP(1) - 1.687500 * RNAT(ITR, IGP) + 1.687500 DERSHP(4) = DERSHP(1) - 1.687500 * RNAT(ITR, IGP) + 1.687500 DERSHP(2) = DERSHP(2) - 1.687500 * RNAT(ITR, IGP) + 2.062500 * RNAT(ITR, IGP) - 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.5500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.687500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.5500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.5500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.5500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.5500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.5500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) + 1.5500 * RNAT(ITR, IGP) + 2.00 DERSHP(2) = DERSHP(2) = 0.5500 * RNAT(ITR, IGP) +

RRBC(NONPDF) = Y(NONPDF)/RKBC(NONPDF,MAXHF) DO 70 J=I+1,MIN(NONPDF,INCRI+2*MAXHF-1) SUM1 = SUM1 + RKBC(I,J-INCRI)*RRBC(J) RRBC(I) = (Y(I) - SUM1)/RKBC(I,MAXHF) CONTINUE 33333 1 EEEEEE EEEEEE EEEEEE EEEEEE EEEEE н Perform backward substitution. NA NA NA ₹₹ ŝ Ē Y(I) = RRBC(I) - SUM1 MMMMMMM ŧ ₹ DO 60 I-NONPDF-1,1,-1 ŧ ξĒ INCRI = I-MAXHF ξĒ £ INCRI = I-MAXHF 40 I=2, NONPDF SUM1 = 0.00 \$\$\$\$\$\$\$5 \$\$ SUM1 = 0.00 CONTINUE \$\$\$\$\$\$\$ \$\$\$\$\$\$ \$\$\$\$\$\$\$ \$\$\$\$\$\$ CONTINUE return End 8 ß \$ 20 8 IF (I.NE.J.AND.INDEX1.GT.0.AND.INDEX1.LE.MAXBW) THEN
RKBC(J,INDEX1) = (RKBCCJ,INDEX1)-SUM1)/RKBC(I,MAXHF) IF (IMDEX1.GT.0.AMD.INDEX1.LE.MAX8W.AMD. INDEX2.GT.0.AND.INDEX2.LE.MAX8W) SUM1 = SUM1+RKBC(J,INDEX1)*RKBC(M,INDEX2) IF (INDEX3.GT.0.AND.INDEX3.LE.MAX8W.AND. INDEX4.GT.0.AND.INDEX4.LE.MAX8W) SUM2 = SUM2+RKBC(I,INDEX3)*RKBC(M,INDEX4) IF (INDEX2.GT.0.AND.INDEX2.LE.MAXBW) RKBC(I,INDEX2) - RKBC(I,INDEX2) - SUM2 ***** +----INDEX2 - I-INCRM INDEX1 = M-INCRJ INDEX3 = M-INCRI INDEX4 = J-INCRM Perform forward elimination. INCRM - M-MAXHF INDEX1 = I-INCRJ INDEX2 = J-INCRI INCRI = I-MAXHF DO 30 M-1,I-1 INDEX1 = 1-INCRJ SUM1 = 0.00 SUM2 = 0.00 INCRU = J-MAXHF CONTENUE DO 10 J=2, NONPDF C 20 I=1,J ENDIF CONT INUE CONTINUE + + + + + 8 19 20

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DO 50 J=MAX(1,INCRI),I-1 IF (J.GT.INCRI) SUM1 + RKBC(I,J-INCRI)*Y(J) CONTINUE SUBROUTINE SWELL(IPRD, LINC, NLINC, PCTREL, PCTSWL, ATMREL) וווווו רדרדרד -----1 רררררר רווווו

Y(1) = RRBC(1)

+4.00+PI+CONCIP(N)/3.00+RADIP(N)++3+MMULT VELCUM(N) = VELCUM(N) + VELIP(N)+MMULT VELIP(N) = 3.00 / (BLTZMM*(TEMP(1)**2)*(LTPARM**3))
*DIFFS*TMPGR(1)*PI*(RADIP(N)**3)*SRFACT + Calculate numerical integration multipliers times R**2 ATOM(N) = ATOM(N) + CONCIP(N)*QFLOAT(N)*MMULT +----= SHPFCT(1)*RADIUS(NODE1,N) +
SHPFCT(2)*RADIUS(NODE2,N) +
SHPFCT(3)*RADIUS(NODE3,N) MMULT = 4.Q0*PI*WEIGHT(ITR,IGP)*RGP**2*AJDET +------+ CONCIP(N) = SHPFCT(1)*CONC(ND0F1)+ SHPFCT(2)*CONC(ND0F2)+ SHPFCT(3)*CONC(ND0F3) Calculate Jacobian matrix and determinant NDOF3 = ICONDF(IELEM, N+2*NBBLSZ) AJMAT = DERSHP(1) * R(ICON(IELEM,1)) + DERSHP(2) * R(ICON(IELEM,2)) + DERSHP(3) * R(ICON(IELEM,3)) NDOF1 = ICONDF(IELEM,N) NDOF2 = ICONDF(IELEM,N+NBBLSZ) . VOLCUM(N) = VOLCUM(N) = SHPFCT(1)*R(NODE1)+ SHPFCT(2)*R(NODE2)+ SHPFCT(3)*R(NODE3) +----DO 101 N=1, NBBLSZ * Loop over matrix rows ******************* RADIP(N) AJDET = AJMAT ----* RGP * + + + + + + + + + + +---- Value of R in real coordinates of integration point Zero shape factor and shape derivitive matricis + +-----CALL SHAPE(NNELEM, ITR, IGP, SHPFCT, DERSHP) VOLUME = 0.00 GRNVOL = 4.00*PI*GRNDIA**3/3.00 DO 151 N=1,NBBLSZ ATOM(N) = 0.00 Loop over integration points | +----*-----NODE1 = ICON(IELEM,1) NODE2 = ICON(IELEM,2) NODE3 = ICON(IELEM,3) * Get global node numbers vol.cum(n) = 0.00 Vel.cum(n) = 0.00 SHPFCT(1) = 0.000 SHPFCT(2) = 0.000 SHPFCT(3) = 0.000 INCLUDE (LATTICE) INCLUDE (BOLTZMAN) INCLUDE (SURFENGY) DO 100 IELEM-1,NE DO 701 ITR=1, IGP (GAUSS) OLDVOL = VOLUME (MAIN) (ELKF) (Id) ATOMS = 0.00 CONTINUE INCLUDE INCLUDE ENCLUDE **INCLUDE** * -----151

PCTSML = 100.00*VOLUME/GRNVOL PCTSML = QMAX1(PCTSML,0.00) PCTSML = QMAX1(PCTSML,1.02) ATMREL = QMAX1(CTOTATM-ATOMS),ATMREL) PCTREL = 100.00*ATMREL/TOTATM PCTREL = QMAX1(PCTREL,0.00) PCTREL = QMIN1(PCTREL,1.02) CONTINUE CONTINUE RETURN END 1031 1030 PCTKPT = (16.Q0*GRNDIA**3 -12.Q0*5HIFT*GRNDIA**2+5HIFT**3)/(16.Q0*GRNDIA**3) ATOMS = ATOMS + ATOM(1))*ATMKPT*PCTKPT VOLUME = VOLUME + VOLCUM(1))*AIMKPT*PCTKPT DO 1031 IK = 1,NN WTMOLE = ENRICH*235.043928Q0 + (1.00-ENRICH)*238.050788Q0 BUMMTL = BUMMTL + 1.394Q19*0MEGA*POWER(1)*DELTIM(IPRD)/WTMOLE DELCON(INDEX) = DELCON(INDEX)*ATMKPT*PCTKPT POWSLW = POWSLW+(POWER1(I,1)+POWER2(I,1)
+(POWER2(I,1)-POWER1(I,1))
/QFLOAT(NLINC(I)))
*(ENDITM(I)-ENDITM(I-1))/2.Q0 INDEX = (IK-1)*NBBLSZ+IJ CONC(INDEX) = CONC(INDEX)*ATMKPT*PCTKPT IF (IREST.EQ.1) TOTATM = TOTATM+RESATM ATWKPT = QMINI((TOTATM/ATOMS),1.Q0) BURATE = 1.72800Q7+FISSRC(1)*OMEGA BUAPPR = BURATE*DELTIM(IPRD)/8.6400Q4 BUAPTL = BUAPTL + BUAPPR TOTATM = POWSUM * 7.80275Q9 * GRNVOL SHIFT - AVGVEL * DELTIM(IPRD) PERPOW = POWER(1)*DELTIM(IPRD) AVGVEL - VELCUM(I)/GRNVOL DO 1032 IJ = 1, NBBLSZ ATOMS = ATOMS + ATOM(IJ) ATOMS = 0.00 DO 1030 I) = 1,NBBLSZ POWSUM = POWSUM+PERPOW IF (LINC.EQ.1) THEN POWSUM = 0.Q0 DO 800 I=1,IPRD-1 **CONTINUE** CONTINUE CONTINUE CONTINUE CONTINUE ENDIF ÷ + + 701 100 1032 101

A2.1 Include Files				, , ₩ ₹	 number of elements NOTE: Problem set up f number of nodes
MANANAN	AAAAA IIIIIIII	NZ	NN	• • •	NOTE: Prodiem set up r (4 local modes f NSPC - number of specified
	AAAA		~	•	NOTE: Only first and l
	AA AA II AA AA II	NN NNNN NN		*	concentrations.
E₹	AAAA	NNNN NN			
₹		NN	NNN	+ NELT	
	AA		N	•	NOTE: Based only on nu
	AA AA IIIIIII	Ž	N	* NNE	NNELEM - number of nodes pe NOTE: Restricted to 2
				NONI + +	INONPD - number of non-pres
IMPLICIT REAL*16 (A-H, 0-Z)	Н,0-Z)			• •	for bubbles
TMPLICII INIEGEN (I-N)	(N			NONI +	NUIE: SIMPLY NUOF MINU INONPV - number of non-pres
COMMON/STORAG/R, CONC, DELCON, RAD, DELRAD, RADIUS, DIFFCF, VOLCUM	DELCON, RAD, DELRAD,	RADIUS, D	IFFCF, VOLCUM	•	for vacancies
+ , BURATE, BUAPPI	BURATE, BUAPPR, BUAPTL, BUANTL, RESOLP, RESADJ, RADMAX, RADMIN	JLP, RESA	DJ, RADMAX, RADMIN	•	NOTE: Simply (NN-1)*2
+ , CONCV, CONCI, I	CONCV, CONCI, DELCNV, DELCNI, BBLPRS, CONCVE, ATOM, VELCUM	S, CONCVE	, ATOM, VELCUM	* MAXH	WAXH - the maximum half ban
+ , DENSTY, POROS	DENSTY, POROS, VOL, SWEL, GRNDIA, SPDISP, DIFFS, YPRS, ENRICH	DISP, DIF	FS, YPRS, ENRICH	•	for bubbles
+ , LIPAKM, UMEGA	LIPAKM,UMEGA,LAMBUA,ZVAL,ZINI,UIFFN,UIFFU,UIFFV,UIFFI DAWED1 DAWED1 TEUD1 TEUD1 DEESS1 DEESS1 TUDED1 TUDED2	LFFN,UIF	FU,UIFFV,UIFFI TWDC01 TWDC03	+ MAAH	MAXHV - THE MAXIMUM NALT DA Ean vicconting
+ , POWERL, POWER	FUNEKL/FUNEKC/IEMPL/IEMPC/FKESSL/FKESSC/IMPUKL/IMP Dowed temd ddecc tuded ddenen eicede Atmede Vafede	1, FRE332	, IMPGKL, IMPGKL	* NEDO	TUP Vacancies No – number of degrees
+ , DELTIM, ENDTIN	DELTIM. ENDTIM. POWSUM. RESATM. TOTATM	VTM	MONL, VALONC		NEDUT - RUMBER UN GEBREGS UN NOTE: Simply NNELEM*NB
+ , , RKBC, RRBC, ELK, ELF	K, ELF			+ NBBL	NBBLSZ - number of bubble g
+ , NE, NN, NSPC, N	NE, NN, NSPC, NMAT, NNELEM, NBBLSZ, NPRD	QX QX		•	NOTE: Currently limite
+ , , NNETYP, NEDOF, NDOF	, NDOF			*	The bubble group
+ , ICON, ICONDF				•	of gas atoms in
+ JARRAY,LARRAY	~			+ NDOF	NDOF - number of degrees of
+ , INONPD, IDOFPI	INONPD, IDOFPD, INONPV, INONPR			*	NOTE: Simply NN*NBBLSZ
+ , IPRBTP, IMAT, I	IPRBTP, IMAT, NPD, IDIR, IREST			* NPRD) - number of time perio
+ , , MAXH, IGP, MAXHV, MAXHR	HV, MAXHR			•	NOTE: Currently limite
+ HEADER				+ INI	INTEGER*4 NE. NN. NSPC. NMAT.
				+	, INONPD, MAXH, NEDO
Header variables.					li i) <u>anning of</u> al
CHARACTER*80 HEADER(10)	10)				to local node i
	•			•	Dim: i = max number
******************************** *				•	j = max number

NE - number of elements NOTE: Problem set up for maximum of 10 elements
NN - number of nodes NOTE: Problem set up for maximum of 21 global nodes
(4 local nodes for 10 elements) NSPC - number of specified concentrations
NMAI - number of material types NOTF furrently only one moterial type ollowed
NELT - number of element types
NOTE: Based only on number of nodes per element
NNELEM - number of nodes per element NOTE: Destricted to 2 or 3
INONPD - number of non-prescribed degrees of freedom
for bubbles
LNUNFY - NUMBER OF NON-PRESCRIDED DEGREES OF TREEDOM for vorancias
NOTE: Simply (NN-1)*2
MAXH - the maximum half bandwidth of stiffness matrix
TOP DUDDIES WAXHV - the moximum half bandwidth of stiffness matrix
NEDOF - number of degrees of freedom per element
NOTE: Simply NNELEM*NBBLSZ
NBBLSZ - number of bubble groups per node NOTF: Currently limited to 50 hubble arouns
The bubble group index determines the number
of gas atoms in the bubble
NDOF - number of degrees of freedom
_
NTKU - number of time periods NOTE: Currently limited to 10
INTEGER*4 NE, NN, NSPC, MMAT, NELT, MNELEM
+ , , , , , , , , , , , , , , , , , , ,
ICON(i,j) - mapping of global node number to local node j of element i
Dim: i = max number of global nodes

Integer variables.

INTEGER#4 ICON(21,4)

	THIEREN & TECHNERT +)
2 -	NLINC(i) - number of time steps in period i Dim: i = max number of períods LOUT(i) - time steps per printout in period i Dim: i = max number of periods
	INTEGER*4 NLINC(10) INTEGER*4 LOUT(10)
	IMAT(i) - type of material for element i NOTE: Currently set to 1 for all elements Dim: i = max number of elements
	INTEGER*4 IMAT(10)
	<pre>NPD(i) - node for prescribed concentration i Dim: i = max number of prescribed concentrations IDIR(i) - bubble group for prescribed concentration i Dim: i = max number of prescribed concentrations</pre>
	INTEGER*4 NPD(50) INTEGER*4 IDIR(50)
i iii	<pre>ICONDF(i,j) - mapping of global degrees of freedom to local degrees of freedom (i.e. bubble groul Dim: i = max number of elements Dim: i = max number of degrees of freedom / element</pre>
	ICONDF(10,150)
	JARRAY(i) - mapping of non-prescribed degrees of freedom to row i of global matrix Dim: i = max number of global degrees of freedom LARRAY(i) - mapping of global matrix rows to degree of freedom i Dim: i = max number of global degrees of freedom
	INTEGER*4 JARRAY(1050) INTEGER*4 JARRAY(1050) INTEGER*4 LARRAY(1050)
a	<pre>IDOFPD(i) - connectivity array for prescribed degree of freedom Dim: i = max number of prescribed concentrations</pre>
1111	

INTEGER*4 IDOFPD(50)

214

 VOLCUM(i) - integrated volumes for i'th bubble group Dim: i = max number of bubble groups 	+ TMPGR
REAL*16 RADWAX(50), RADWIN(50), ATOM(50), VELCUM(50),VOLCUM(50)	
اس ا	PRSNG
 DUMER2(i,j) = max number or materials POWER2(i,j) - thermal final power of i'th period and i'th material 	FISSR
Dim: i = max number of periods Dim: i = max number of materials	* ATMSRI *
 	* VACSRI
Dim: i = max number of periods	* CONCV
	• • •
 Uim: 1 = max number of periods Dim: j = max number of materials TEMPI(i,j) - initial temperature of i'th period And i'th material material 	REAL
 Dim: i = max number of periods Dim: j = max number of materials TEMP2(i,j) - final temperature of i'th period 	blffC
 Dim: i = max number of periods Dim: j = max number of materials TNPGR1(i,j) - initial temperature gradient of i'th period 	RADIU
ب المعامي	88LPR
Dim: j = max number of periods	REAL
<pre>REAL*16 POWER1(10,1), POWER2(10,1) + , TEMP1(10,1), TEMP2(10,1) + , TMPGR1(10,1), TMPGR2(10,1) + , PRESS1(10,1), PRESS2(10,1)</pre>	DELCO
 POWER(i) - thermal power for i'th material Dim: i = max number of materials 	* * RAD(1)

L*16 POMER(1), TEMP(1), PRESS(1), PRSNGN(1), FISSRC(1), ATMSRC(1)
L*16 TMPGR(1), VACSRC(1), CONCVE(1,2) equilibrium interstitial (j=1) concentrations| Dim: i = max number of materials M(i) - change in concentration since last time step Dim: i = max number of global degree of freedom i) - vacancy concentration at end of last time step Dim: i = max number of global degree of freedom XC(i) - gas atom production rate for i'th material XC(i) - vacancy production rate for i'th material *16 DIFFCF(21,50), RADIUS(21,50), BBLPRS(21,50) XC(i) - fission rate density for i'th material Cf(i,j) - diffusion coefficient for i'th node S(i) - hydrostatic pressure for i'th material X(i) - temperature gradient for i'th material (i) - concentration at end of last time step for i'th global degree of freedom for i'th global degree of freedom /E(i,j) - equilibrium vacancy (j=2) and max number of bubble groups = max number of bubble groups S(i,j) - bubble pressure for i'th node Dim: j = max number of bubble groups Dim: j = max number v. US(i,j) - bubble radius for i'th node¹ -i'th hubble group SN(i) - partial pressure of nitrogen Dim: i = max number of materials (i) - temperature for i'th material and j'th bubble group Dim: i = max number of nodes and j'th bubble group = max number of nodes and j'th bubble group Dim: i = max number of nodes ----***************** set to 1 Dim: j = 2 Dim: i Dim: j NOTE:

Dim: i = max number of global degree of freedom	REAL*16 ELK(150,1
<pre>DELRAD(i) - change in concentration since last time step for i'th global degree of freedom</pre>	 ELK Dimensions
Dim: i = max number of global degree of freedom CONCV(i) - vacancy concentration at end of last time step	DATA IDMEK1, IDME
	<pre>* ELF Dimensions</pre>
	DATA IDMEF1,IDME
Dim: i = max number of nodes CONCI(i) - interstitial concentration at end of last time	* ENDTIM(i) - end t
step for i'th mode	* Dim: i = m
Dim: i = max number of nodes DELCNI(i) - change in concentration since last time step	* NOTE: '8'
for i'th node Dim: i = max number of nodes	KEALTIG ENULIA(0
<pre>REAL*16 CONC(1050),DELCON(1050) REAL*16 RAD(1050),DELRAD(1050) REAL*16 RAD(1050),DELRAD(1050) REAL*16 CONC(21),DELCNV(21) REAL*16 CONCI(21),DELCNI(21)</pre>	00000
<pre>SPDISP(i) - value of specified concentration i Dim: i = max number of prescribed concentrations 1</pre>	888
REAL*16 SPDISP(50)	*88888888 0000000 FFF1 *88888888 0000000 FFF1
RKBC(i,j) - stiffness matrix [t i = max number of global degrees of freedom Dim: j = max number of local degrees of freedom	* REAL*16 BLTZWN
. • •	
REAL*16 RKBC(1050,300), RRBC(1050)	
RBC Dimensions I	
DATA IDWRB1, IDWRB2/1050, 300/	
<pre>ELK(i,j) - elemental stiffness matrix ELK(i,j) - elemental stiffness matrix Dim: i = max number of local degrees of freedom ELE(i) _ elemental force variant</pre>	**************************************

REAL	L*16 ELK(REAL*16 ELK(150,150),ELF(150)	:LF(150)				+	
ELK Dir	ELK Dimensions	+ ·						
DATA	A IDMEK1,	DATA IDMEK1, IDMEK2, IDMEK3/150, 150, 1/	AEK3/150,1	11/				
ELF Dir	ELF Dimensions	+						
DAT	: E		ÆF3/1,150	,1/				
ENDT	ENDTIM(i) - e Dim: i NOTE:	p so	time for period i max number of peri index needed for	d i in real periods for initial	time time	(seconds) period	• ·	
REAL	REAL*16 ENDTIM(0:10)	IM(0:10)		, , , ,		1 1 1 1 1 1 1	+	
38888888 38888888 38888888 38888888 3888888	000000 0000000 0000000 0000000 00000000	ווווווו ווווווו ורוווווו ור		22222222 22222222 2222 222 222 222 222		Ададада Ададада Ададада Ададада Ададада Ададада Ададада Ададада Ададада	NN NN NNN NN NNN NN NNN NN NNN NN NNN NN NNN NN NNN NN NNN NN NNN NN	
REAL Units o	REAL*16 BLTZMN Units are in erg/K	MN + g/K		9 			+	
DATA	A BLTZMN/	DATA BLTZMN/1.380662Q-16/	16/					

<pre>Dim: j = max number of local degrees of freedom TELF(i,j) - temporary elemental force vector Dim: i = max number of local degrees of freedom REAL*16 TELK(150,150), TELF(150) REAL*16 CONCIP(50), DELCIP(50), DERCIP(50), SRCFRC(50) REAL*16 CONCIP(50), DELCIP(50), DERCIP(50), VELIP(50), RESOL(50) REAL*16 RADIP(50), DFCIP(50), BPRSIP(50), VELIP(50), RESOL(50) REAL*16 BPRSA(50), BPRSB(50), BPTMIP(50) REAL*16 BPRSA(50), BPRSB(50), BPRSB(50), BPTMIP(50) REAL*16 BPRSA(50), BPRSB(50), BPRSB(50),</pre>	GGGGGG AAAAA UU UU SSSSS SSSSSS SSSSSS SSSSSS SSSSSSS SSSSSSSS SSSSSSS SSSSSSS SSSSSSS SSSSSSS SSSSSSS SSSSSSS SSSSSSSS SSSSSSS SSSSSSS	 SHPFCT(i) - shape factor array Dim: i = max number of local nodes DERSHP(i) - shape factor array Dim: i = max number of local nodes 	REAL*16 SHPFCT(4),DERSHP(4)	 RNAT(i,j) - Gauss quadrature roots Dim: i = max number of choices for integration pts WEIGHT(i,j) - Gauss quadrature weights Dim: i = max number of choices for integration pts 	REAL*16 RNAT(6,6),WEIGHT(6,6)
 DIFFVA = pre-exponential for vacancy diffusion coefficient (cm2/s) DIFFIA = pre-exponential for interst diffusion coefficient (ergs*K/ DIFFIB = exponential term for vacancy diffusion coefficient (ergs*K/ DIFFIB = exponential for interst diffusion coefficient (ergs*K/ DIFFXB = exponential for xenon diffusion coefficient (ergs*K/ DIFFXB = exponential for surface diffusion coefficient (ergs*K/ DIFFXB = exponential term for variate diffusion coefficient (ergs*K/ DIFFXB = exponential term for surface diffusion coefficient (ergs*K/ DIFFXB = exponential for uranium self diffusion (cm2/s) DIFFUM = mean jump distance of surface atom (cm) DIFFUM = partial pressure exponential DIFFUM = partial term for uranium self diffusion (cm2/s) DIFFUM = partial pressure exponential DIFFUM = partial press	 REAL*16 DIFFXA, DIFFXB, DIFFSA, DIFFSB REAL*16 DIFFXA, DIFFXB, DIFFSA, DIFFSB DATA DIFFXA, DIFFTB/11, Q-3, Z. 32Q3/ DATA DIFFXA, DIFFXB/4, 65360-04, 3, 6073Q4/ DATA DIFFXA, DIFFSB/11, 92000+03, 6, 5188Q4/ DATA DIFFE/R, DIFFSB/11, 12Q-6/ DATA DIFFE/R, 22Q-31/ DATA DIFFLA, DIFFNB, DIFFNC/22, 2146Q-11, 0, 6414Q0, 7, 9893Q3/ DATA DIFFUA, DIFFNB, DIFFNC/22, 251Q-05, 0, 4134Q0, 1, 92147Q4/ DATA DIFFNA, DIFFNC/22, 251Q-05, 0, 4134Q0, 1, 92147Q4/ DATA DIFFNA, DIFFNOZ/-0, 73700, 2, 1790-4/ 			EEEEEE LL KKKKK FFFFF EEEEEEE LL KKKKK FFFFF EEEEEEEE LLL KK KK FF EEEEEEEE LLLLILL KK KK FF	 TELK(i,j) - temporary elemental stiffness matrix Dim: i = max number of local degrees of freedom

,

0.900, 0.57735026918962600, 0.57735026918962600, 0.57735026918962600, 0.57735026918962600, 0.	TI UUUUUU 00 00 NN NN III *
	<pre>* INTEGER*4 IFLD(40),IDATE(14) LOGICAL ITEND</pre>
-0.861136311594053(00, -0.339981043584856(00, 0.339981043584856(20, 0.861136311594053(0, 0.000.	CHARACTER*80 BUFFER CHARACTER*50 PROBLM(5) CHARACTFR*13 TEMPTP(5)
-0.53846931010	
0.90617984593866400. 0.5384595101050617984593866400.	CHARACTER*10 UNITS(0:8.2),CRDSYS(3,5),CRDUNT(3,2,5) CHARACTER*10 UNITS(0:8.2),CRDSYS(3,5),CRDUNT(3,2,5)
-0.661209386466	CHARACTER*23 DATE
-0236019186083197(%) 0.236019186063197(%) 0.661209336466265(%) 0.932469514203152(%)	CHARACTER*36 MONTHS CHARACTER*36 MONTHS
Gauss quadrature weights	-+ CHARACLER*63 UATS CHARACTER*4 AMPM
DATA WEIGHT/ 0.000, 0.000,	.+ DATA DATE/' '/
	MONTHS/'JanfebMarAprMayJunJulA
1.000, 1.000,	DAYS(1:36)/'Sunday Monday Tu
0.000, 0.000 0.000 0.000	DATA DAYS(37:63) /'Thursday Friday Saturday '/ DATA ANDM/'AMDM''
0.888888888888888	
	DATA PROBLM/'Single Fuel Grain
	+
0.34785484513745400, 0.65214515486254600, 0.65214515486254600. 0.34785484513745400.	
	-+
0.47862867	DATA COORD/' (1=R)
0.56888888888888900, 0.47862867049936600, 0.23695688688888900, 0.47862867049936600, 0.000	
. 0.36076157304	- -
0.46791393	+
0.360761573048139Q0, 0.171324492379170Q0/	DATA ELEMNT/'Spherical ',

IIIIIII II dd dd III dddddd Ad Addddd Ad Addddd Ad A A A A A	* REAL*16 PI DATA PI/3.14159265358979323846264338327950288Q0/	 SSSSSS UU UU RRRRRRR FFFFFFF EEEEEEE NN NN GGGGGGG YY YY SSSSSSS UU UU RRRRRRR FFFFFFF EEEEEEE NN NN GGGGGGGG YY YY SSSSSSS UU UU RRRRRR FFFFF EEEEEE NN NN GG GGGGG YY YY SSSSSSS UU UU RRRRRRR FFFFF EEEEE NN NN GG GGG YY YY SSSSSSS UU UU RRRRRR FFFFF EEEEE NN NN GG GGG YY YY SSSSSSS UU UU RRRRRR FFFF EEEEE NN NN GG GGG YY YY SSSSSSS UUUUUUU RR RF FF EEEEEE NN NN GG GGG YY Y SSSSSSS UUUUUUU RR FF EEEEEEEN NN GGGGGGG YY SSSSSSS UUUUUUU RR FF EEEEEEEN NN GGGGGGG YY 	REAL*16 SRFENG, SRFACT, RESLMH, RESLML Units are in dyn/cm equal to 1 erg SRFACT is activation energy for surface atoms to break free in units of ergs RESLML are high temp and low temp resolution FACTOR STANDIA With terms is defined as 51500 K		VV AAAAAA NN NN DDDDDDD WW WW AAAAAA AAAAAAA VV AAAAAAA NNN NN DDDDDDDD WW WW AAAAAAAA AAAAAAAA	
<pre> tata UNITS/'SI ', m', N/m**2 ', m/m-C ', meters', N ','C',</pre>		DATA CRDUNT/meters', meters', meters', + 'inches', 'inches', 'inches', + 'inches', 'inches', 'radian', + 'inches', 'inches', 'radian', + 'inches', 'inches', 'radian', + 'inches', 'inches', 'inches', + 'inches', 'inches', 'inches',	LL AAAAAA TTTTTTT TTTTTTTT TTTTTTTTTTTT	REAL*16 LTPRMB, LTPRMC REAL*16 EFORMV, DISDEN, DISRAD REAL*16 YPSLPE, YPINT REAL*16 YPSLPE, YPINT	DATA LTPRMA,LTPRMB,LTPRMC/4.8790-8,3.2640-13,6.889Q-17/ DATA EFORMV,DISDEN,DISRAD/1.96Q4,1.Q8,25.Q-8/ DATA YPSLPE,YPINT/-4.6281961549655164Q6,8.6732683308300463Q9/	II dd dd IIIIIIII ddddddd

• Units are in (1) - g cm4 • (2) - cm3	REAL	*16 V	REAL*16 VANDWL(2)		¥ ¥	E	*	W W	A	• VV AA AA NN NN DDDDDDD WW MA AA AA AA AA LLLLLLL REAL*16 VANDML(2)
		rein		Т Т с						

•

APPENDIX 2

SAMPLE INPUT

* OUPUT DECK ? - 0: NO 1: YES 1 000 0 *_____ _____ * HEADER CARDS, MAX 80 CHARACTERS | *-----100 UN Fuel Grain Modelling Using Nonlinear Finite Element Analysis 101 By Capt. Daniel L. DeForest, USAF 102 103 Sample Input Deck ***** 105 ******* CONDITIONS 106 ***** 107 Power Density = 700.00 W/cm3 108 109 Temp = 1200 K110 Temp Grad = 1000 K/cm 111 External Pressure = 1033 Pa Time Length = 1.0E5 s 112 *-----* TYPE OF UNITS | *----+ 201 2 *_____ * NUMBER OF GUASS INTEGRATION POINTS | *------202 3 *_...... * LIST INPUT DECK FLAG T *_____ 203 0 *_____ * RESTART FLAG *------204 0 *_____ * NE, NBBLSZ, NNELEM, NSPC, NPRD, MAXITER | *_____ 300 10 10 3 10 2 50 * R COORDINATES IN ORDER OF NODE *_____ 501 0.000 1.25Q-4 502 2.50-4 503 504 3.50-4 505 4.50-4 506 5.5Q-4 507 6.5Q-4 7.0Q-4 508 509 7.50-4 510 8.00-4 511 8.5Q-4

512 8.90-4 513 9.3Q-4 514 9.50-4 9.70-4 515 9.80-4 516 517 9.90-4 518 9.9250-4 519 9.950-4 520 9.98750-4 521 10.0Q-4 * SPECIFIED CONCENTRATIONS - NODE #, BBLGP, VALUE | *-----601 21 1 0.000 ź 602 21 0.000 3 603 21 0.0Q0 604 21 4 0.000 605 21 5 0.0Q0 606 21 6 0.000 607 21 7 0.000 608 21 8 0.0Q0 609 21 9 0.0Q0 0.000 610 21 10 *-----* GRAIN RADIUS | *-----+ 701 10.Q-4 *_____ * INITIAL POROSITY, INITIAL ENRICHMENT, CONVERGENCE LIMIT | 901 0.057 0.10 1.Q-4 * PERIOD LENGTH, #TIME STEPS, PRINTOUT INTERVAL | *_____ 1001 1.0Q5 5 5 *_____ * MATERIALS - POWER, TEMP, TEMP GRADIENT, PRESSURE | *_____ 1102 700.0 1200.0 1000.0 1033.23 700.0 1200.0 1000.0 1033.23

APPENDIX 3

SAMPLE OUTPUT

•

RRRRRR RRRRRRR	EEEEEEEE		SSSSSS SSSSSSSS		000000	NN NN NNN NN	EEEEEEEE
RR RR	EE	DD DD	SS	Π	00 00	NNNN NN	EE
RR RR	EEEEEE	DD DD	SSSSSSS	Π	00 00	NN NN NN	EEEEEE
RRRRRRR	EEEEEE	DD DD	SSSSSSS	11	00 00	NN NNNN	EEEEEE
RRRRRR	EE	DD DD	SS	π	00 00	NN NNN	EE
RR RR	EEEEEEEE	DDDDDDDD	SSSSSSSS	Π	00000000	NN NN	EEEEEEEE
RR RR	EEEEEEE	DDDDDDD	SSSSSS	Π	000000	NN NN	EEEEEEEE

------ PROGRAM REDSTONE -----

UN Fuel Grain Modelling Using Nonlinear Finite Element Analysis By Capt. Daniel L. DeForest, USAF

> Sample Input Deck CONDITIONS Power Density = 700.00 W/cm3 Temp = 1200 K Temp Grad = 1000 K/cm External Pressure = 1033 Pa Time Length = 1.0E5 s

1 10 Total number of elements Total number of nodes 21 Maximum bubble groups per node 10 . Total number of material types 1 = Total number of specified displacements = 10 Total number of time periods 2 = Maximum number of iterations 50 ÷ Units are in BRITISH

Number of Guass integration points is 3

Grain Diameter = 1.000E-3 microns

NODAL POSITIONS FROM CENTER

	R(I)
Global Node>I	(inches)
1	0.00000E+0
2	1.250000E-4
3	2.500000E-4
4	3.500000E-4
5	4.500000E-4
6	5.500000E-4
7	6.500000E-4
8	7.000000E-4
9	7.500000E-4
10	8.00000E-4
11	8.500000E-4
12	8.90000E-4
13	9.300000E-4
14	9.500000E-4
15	9.700000E-4
16	9.80000E-4
17	9.90000E-4
18	9.925000E-4
19	9.950000E-4
20	9.987500E-4
21	1.000000E-3

ELEMENT CONNECTIVITY

		Global Nod for the Lo Numbers Be	cal Node	}
Element	1	2	3	4
1	1	2	3	0
2	3	4	5	0
2 3	5	6	5 7	0
4	1 3 5 7 9	8	9	0
5	9	10	11	0
5 6 7	11	12	13	0
7	13	14	15	0
8	15	16	17	0
9	17	18	19	0
10	19	20	21	0
	ELEMENT	MATERIAL	PROPERTI	ES
		Material	Initial	
E	lement>I	Type ID	Porosity	y
-	1	1	5.7000	
	1 2 3	1	5.7000	
	3	1	5.7000	
	4	1	5,7000	
	5	1	5.7000	
	6	1	5.7000	
	7	1	5,7000	
	8	1	5.7000	
	9	î	5.7000	
	10	1	5.7000	
	T A	-	2	

1

+

1

+

1

+

TIME HISTORY CONDITIONS

Time Incr	Initial Power	Initial Temp		tial ess	Fina Powe	-	Final Temp		Final Press
+ <u> </u>	700. 0.000E+0	1.200E+3 0.000E+0 TIME	0.0	33E+3 00E+0 INFORM		ØE+0	1.200 0.000	-	1.033E+3 0.000E+0
	Ti. In			Number Time S			Step onds)		ntout erval
+	1 2		0E+05 0E+00		5 0		 000E+04 000E+05		5 0
1									

SPECIFIED CONCENTRATIONS

Global Node #	Bubble Gp	Value
21	1	0.000E+00
21	2	0.000E+00
21	2 3 4	0.000E+00
21	4	0.000E+00
21	5 6	0.000E+00
21	6	0.000E+00
21	7	0.000E+00
21	8	0.000E+00
21	9	0.000E+00
21	10	0.000E+00
TIME PERIOD 1 Time Step 5 Time Step Size Time in Period	= 1.0000E+5	seconds
New Time Time step represe		
Percent fission		
Percent swelling	=	4.2423E-03%
Burnup rate		atom percent/da
Burnup for perio		
Burnup (total)		
	= 122.80	MWD/MT

Power density	=	700.0	W/cm3
Pressure	=	1033.	Ра
Temperature	=	1200.	K

1

	NODAL C	ONCENTRATIONS	
Bubble	Concentration	Bubble Radius	Bubble Pressure
Group	(cm-3)	(cm)	(Pa)
	Node = 1	Radius = 0.0	00E+00
1	1.907419E+17	2.426457E-08	0.000000E+00
ź	1.152346E+17	3.471774E-08	5.760742E+10
3	2.970138E+16	3.980550E-08	5.024431E+10
4	6.723207E+15	4.386735E-08	4.559199E+10
5	1.396060E+15	4.730536E-08	4.227850E+10
	2.723131E+14	5.031653E-08	3.974836E+10
6			
7	5.061627E+13	5.301388E-08	3.772596E+10
8	9.051949E+12	5.546897E-08	3.605619E+10
9		5.773034E-08	3.464382E+10
10	2.973562E+11	5.983260E-08	3.342659E+10
	Node = 2	Radius = 1.2	50E-04
1	1.907419E+17		0.000000E+00
2	1.152346E+17	3.471774E-08	5.760742E+10
3	2.970138E+16	3.980550E-08	5.024431E+10
4	6.723206E+15	4.386735E-08	4.559199E+10
5	1.396059E+15	4.730536E-08	4.227850E+10
6	2.723131E+14	5.031653E-08	3.974836E+10
7	5.061627E+13	5.301388E-08	3.772596E+10
8	9.051950E+12	5.546897E-08	3.605619E+10
9	1.577365E+12	5.773034E-08	3.464382E+10
	2.973562E+11	5.983260E-08	3.342659E+10
	Node = 3	Padius - 25	00K_0A
	$=$ Noue = $5^{}$	-1 Rualus = 2.5	001-04
1	1.907419E+17	2.426457E-08	0.000000E+00
2	1.152346E+17	3.471774E- 0 8	5.760742E+10
3	2.970138E+16	3.980550E-08	5.024431E+10
4	6.723206E+15	4.386735E-08	4.559199E+10
5	1.396059E+15	4.730536E-08	4.227850E+10
6	2.723131E+14	5.031653E-08	3.974836E+10
7	5.061627E+13	5.301388E-08	3.772596E+10
8	9.051950E+12	5.546897E-08	3.605619E+10
	1.577364E+12	5.773034E-08	3.464382E+10
9 10	2.973562E+11	5.983260E-08	3.342659E+10
	Node = 4	Radius = 3.5	00E-04
1	1.907419E+17	2.426457E-08	0.000000E+00
1 2 3 4	1.152346E+17	3.471774E-08	5.760742E+10
2	2.970138E+16	3.980550E-08	5.024431E+10
С А		4.386735E-08	4.559199E+10
4 r	6.723206E+15		4.227850E+10
5 6	1.396059E+15	4.730536E-08	
6	2.723131E+14	5.031653E-08	3.974836E+10
7	5.061627E+13	5.301388E-08	3.772596E+10
8	9.051950E+12	5.546897E-08	3.605619E+10
9	1.577365E+12	5.773034E-08	3.464382E+10
10	2.973562E+11	5.983260E-08	3.342659E+10
	Node = 5	Radius = 4.5	500E-04
	4 007/105 17	2 4264575 00	0 0000005.00
1	1.907419E+17	2.426457E-08	0.000000E+00
2	1.152346E+17	3.471774E-08	5.760742E+10
3	2.970138E+16	3.980550E-08	5.024431E+10

+

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4	6.723207E+15	4.386735E-08	4.559199E+10	
5	1,396060E+15	4.730536E-08	4.227850E+10	
6	2.723131E+14	5.031653E-08	3.974836E+10	
7	5.061627E+13	5.301388E-08	3.772596E+10	
. 8	9.051949E+12	5.546897E-08	3.605619E+10	
		5.773034E-08		
9	1.577364E+12		3.464382E+10	
10	2.973562E+11	5.983260E-08	3.342659E+10	
	Node = 6	- Radius = 5.5	00E-04	
1	1.907419E+17	2.426457E-08	0.00000E+00	
2	1.152346E+17	3.471774E-08	5.760742E+10	
3	2.970138E+16	3.980550E-08	5.024431E+10	
4	6.723206E+15	4.386735E-08	4.559199E+10	
5	1.396059E+15	4.730536E-08	4.227850E+10	
	2.723131E+14	5.031653E-08	3.974836E+10	
6				
7	5.061627E+13	5.301388E-08	3.772596E+10	
8	9.051950E+12	5.546897E-08	3.605619E+10	
9	1.577365E+12	5.773034E-08	3.464382E+10	
10	2.973563E+11	5.983260E-08	3.342659E+10	
	Node = 7	Radius = 6.5	00E-04	
1	1.907419E+17	2.426457E-08	0.000000E+00	
ź	1.152346E+17	3.471774E-08	5.760742E+10	
3	2.970139E+16	3.980550E-08	5.024431E+10	
4	6.723208E+15	4.386735E-08	4.559199E+10	
5	1.396060E+15	4.730536E-08	4.227850E+10	
6	2.723132E+14	5.031653E-08	3.974836E+10	
7	5.061626E+13	5.301388E-08	3.772596E+10	
8	9.051948E+12	5.546897E-08	3.605619E+10	
9	1.577364E+12	5.773034E-08	3.464382E+10	
10	2.973560E+11	5.983260E-08	3.342659E+10	
	Node = 8	Radius = 7.0	00E-04	
	1 0074405.47	2 4264575 00	0 0000005.00	
1	1.907419E+17	2.426457E-08	0.00000E+00	
2	1.152345E+17	3.471774E-08	5.760742E+10	
3	2.970137E+16	3.980550E-08	5.024431E+10	
4	6.723204E+15	4.386735E-08	4.559199E+10	
5	1.396059E+15	4.730536E-08	4.227850E+10	
6	2.723131E+14	5.031653E-08	3.974836E+10	
7	5.061627E+13	5.301388E-08	3.772596E+10	
8	9.051952E+12	5.546897E-08	3.605619E+10	
9	1.577365E+12	5.773034E-08	3.464382E+10	
	2.973565E+11	5.983260E-08	3.342659E+10	
10	2.9/30050+11	3.3032005-00	3.3420330+10	
Node = 9 Radius = 7.500E-04				
-			D 000007 00	
1	1.907420E+17	2.426457E-08	0.00000E+00	
2	1.152348E+17	3.471774E-08	5.760742E+10	
3	2.970145E+16	3.980550E-08	5.024431E+10	
4	6.723221E+15	4.386735E-08	4.559199E+10	
5	1.396062E+15	4.730536E-08	4.227850E+10	
6	2.723133E+14	5.031653E-08	3.974836E+10	
7	5.061624E+13	5.301388E-08	3.772596E+10	
		5.546897E-08	3.605619E+10	
8	9.051934E+12			
9	1.577360E+12	5.773034E-08	3.464382E+10	
10	2.973548E+11	5.983260E-08	3.342659E+10	

	Node = 10	Radius = 8.0	00E-04	
1	1.907418E+17	2.426457E-08	0.000000E+00	
ź	1.152344E+17	3.471774E-08	5.760742E+10	
3	2.970132E+16	3.980550E-08	5.024431E+10	
4	6.723194E+15	4.386735E-08	4.559199E+10	
5	1.396058E+15	4.730536E-08	4.227850E+10	
6	2.723130E+14	5.031653E-08	3.974836E+10	
7	5.061629E+13	5.301388E-08	3.772596E+10	
8	9.051963E+12	5.546897E-08	3.605619E+10	
9	1.577369E+12	5.773034E-08	3.464382E+10	
10	2.973574E+11	5.983260E-08	3.342659E+10	
	Node = 11	Radius = 8.5	00E-04	
1	1.907426E+17	2.426457E-08	0.000000E+00	
Z	1.152358E+17	3.471774E-08	5.760742E+10	
3	2.970175E+16	3.980550E-08	5.024431E+10	
3 4	6.723284E+15	4.386735E-08	4.559199E+10	
5	1.396071E+15	4.730536E-08	4.227850E+10	
6	2.723141E+14	5.031653E-08	3.974836E+10	
7	5.061614E+13	5.301388E-08	3.772596E+10	
8	9.051868E+12		3.605619E+10	
°9	1.577339E+12	5.773034E-08	3.464382E+10	
		5.983260E-08		
10	2.973490E+11	2.9032005-00	3.342659E+10	
	Node = 12			
1	1.907413E+17	2.426457E-08	0.000000E+00	
2	1.152335E+17	3.471774E-08	5.760742E+10	
3 4	2.970104E+16	3.980550E-08	5.024431E+10	
4	6.723135E+15	4.386735E-08	4.559199E+10	
5	1.396049E+15	4.730536E-08	4.227850E+10	
6	2.723123E+14	5.031653E-08	3.974836E+10	
7	5.061639E+13	5.301388E-08	3.772596E+10	
8	9.052025E+12	5.546897E-08	3.605619E+10	
9	1.577388E+12	5.773034E-08	3.464382E+10	
10	2.973629E+11	5.983260E-08	3.342659E+10	
	Node = 13			
1	1.907460E+17	2.426457E-08	0.00000E+00	
2	1.152416E+17		5.760742E+10	
3	2.970360E+16	3.980550E-08	5.024431E+10	
4	6.723668E+15	4.386735E-08	4.559199E+10	
5	1.396128E+15	4.730536E-08	4.227850E+10	
6	2.723188E+14	5.031653E-08	3.974836E+10	
7	5.061556E+13	5.301388E-08	3.772596E+10	
8	9.051475E+12	5.546897E-08	3.605619E+10	
9	1.577216E+12	5.773034E-08	3.464382E+10	
10	2.973139E+11	5.983260E-08	3.342659E+10	
Node = 14 Radius = 9.500E-04				
1	1.907370E+17	2.426457E-08	0.000000E+00	
2	1.152263E+17	3.471774E-08	5.760742E+10	
3	2.969876E+16	3.980550E-08	5.024431E+10	
4	6.722662E+15	4.386735E-08	4.559199E+10	
5	1.395979E+15	4.730536E-08	4.227850E+10	
6	2.723068E+14	5.031653E~08	3.974836E+10	

7	5.061717E+13	5.301388E-08	> 7775065.10
			3.772596E+10
8	9.052528E+12	5.546897E-08	3.605619E+10
9	1.577544E+12	5.773034E-08	3.464382E+10
10	2.97 40 71E+11	5.983260E-08	3.342659E+10
	Node = 15	Rodius - 97	00F-04
	1006 - 13	$= \pi u u u u = 9.7$	002-04
1	1.907768E+17	2.426457E-08	0.000000E+00
2	1.152929E+17	3.471774E-08	5.760742E+10
2			
3	2.971989E+16	3.980550E-08	5.024431E+10
4	6.727095E+15	4.386735E-08	4.559199E+10
5	1.396649E+15	4.730536E-08	4.227850E+10
5			
6	2.723653E+14	5.031653E-08	3.974836E+10
7	5.061180E+13	5.301388E-08	3.772596E+10
8	9.048326E+12	5.546897E-08	3.605619E+10
9	1.576200E+12	5.773034E-08	3.464382E+10
10	2.970200E+11	5.983260E-08	3.342659E+10
	2.5.02002.22	0.0002002 00	5.5 120552 . 20
	Node = 16	Radius = 9.8	00E-04
	4 0030435 43	2 4264575 00	a aaaaaar aa
1	1.907013E+17	2.426457E-08	0.000000E+00
2	1.151657E+17	3.471774E-08	5.760742E+10
3	2.967965E+16	3.980550E-08	5.024431E+10
4	6.718704E+15	4.386735E-08	4.559199E+10
5	1,395400E+15	4.730536E-08	4.227850E+10
č	2.722627E+14		
6		5.031653E-08	3.974836E+10
7	5.062442E+13	5.301388E-08	3.772596E+10
8	9.056896E+12	5.546897E-08	3.605619E+10
9	1.578883E+12		3.464382E+10
10	2.977862E+11	5.983260E-08	3.342659E+10
	No. 1. 47		005 04
	Node = 17	Kadius = 9.9	00E-04
1	1.910678E+17	2.426457E-08	0.000000E+00
2	1.157624E+17	3.471774E-08	5.760742E+10
3	2.987072E+16	3.980550E-08	5.024431E+10
4	6.759842E+15	4.386735E-08	4.559199E+10
			-
5	1.401990E+15	4.730536E-08	4.227850E+10
6	2.729704E+14	5.031653E-08	3.974836E+10
7	5.062448E+13	5.301388E-08	3.772596E+10
8	9.030537E+12	5.546897E-08	3.605619E+10
9	1.569303E+12	5.773034E-08	3.464382E+10
10	2.948855E+11	5.983260E-08	3.342659E+10
	Node = 18	Deditor 0.0	255 AA
	NOUE = 10	Radius = 9.9	23E-04
1	1 912319F±17	2.426457E-08	0.000000E+00
-			
2	1.154131E+17	3.471774E- 0 8	5.760742E+10
3	2.983335E+16	3.980550E-08	5.024431E+10
4	6.792335E+15		4.559199E+10
5	1.421386E+15	4.730536E-08 5.031653E-08	4.227850E+10
6	2.797620E+14	5.031653E-08	3.974836E+10
	5 251272E-13		
7		5.301388E-08	
8	0 A00107E.17	5.546897E-08	3.605619E+10
•	9.488197E+12	0.0.000012 00	
	9.466197E+12 1 671186F+12	5 773034F-08	3 464382F+10
9	1.671186E+12	5.773034E-08	
	1.671186E+12	5.983260E-08	
9	1.671186E+12	5.773034E-08	
9 10	1.671186E+12 3.188721E+11	5.773034E-08 5.983260E-08	3.342659E+10
9 10	1.671186E+12	5.773034E-08 5.983260E-08	3.342659E+10
9 10	1.671186E+12 3.188721E+11 Node = 19	5.773034E-08 5.983260E-08 Radius = 9.9	3.342659E+10 50E-04
9 10	1.671186E+12 3.188721E+11 Node = 19	5.773034E-08 5.983260E-08	3.342659E+10 50E-04

2	1.291832E+17	3.471774E-08	5.760742E+10
3	3.460673E+16	3.980550E-08	5.024431E+10
4	8.009758E+15	4.386735E-08	4.559199E+10
5	1.682402E+15	4.730536E-08	4.227850E+10
6	3.296526E+14	5.031653E-08	3.974836E+10
7	6.127661E+13	5.301388E-08	3.772596E+10
8	1.092718E+13		3.605619E+10
9	1.895029E+12	5.773034E-08	3.464382E+10
10	3.547876E+11	5.983260E-08	3.342659E+10
10	3.34/0/00+11	3.963200E-08	3.3420396+10
	No. 4		075 04
	Node = 20	Kadius = 9.9	87E-04
1	1.107092E+17	2.426457E-08	0.000000E+00
ź	4,447289E+16	3.471774E-08	5.760742E+10
3	7.309087E+15	3.980550E-08	5.024431E+10
4	1.026271E+15	4.386735E-08	4.559199E+10
5	1.281519E+14	4.730536E-08	4.227850E+10
6	1.434489E+13	5.031653E-08	3.974836E+10
7	1.403750E+12	5.301388E-08	3.772596E+10
8	1.078410E+11	5.546897E-08	3.605619E+10
9	3.013661E+09	5.773034E-08	3.464382E+10
10	0.000000E+00	5.983260E-08	3.342659E+10
-			
	Node = 21	Radius = 1.0	00E-03
1	0.000000E+00	2.426457E-08	0.000000E+00
2	0.000000E+00	3.471774E-08	5.760742E+10
2	0.000000E+00 0.000000E+00	3.980550E-08	5.024431E+10
4			
	0.000000E+00	4.386735E-08	4.559199E+10
5	0.000000E+00	4.730536E-08	4.227850E+10
6	0.000000E+00	5.031653E-08	3.974836E+10
7	0.000000E+00	5.301388E-08	3.772596E+10
8	0.000000E+00	5.546897E-08	3.605619E+10
9	0.00000E+00	5.773034E-08	3.464382E+10
10	0.000000E+00	5.983260E-08	3.342659E+10

VITA

Daniel Lee DeForest is a Captain in the United States Air Force. He first joined the Air Force in 1980 soon after completing his undergraduate work. His undergraduate degree was a Bachelor of Science in Chemistry Cum Laude received from Baylor University in 1980. He subsequently received a Masters of Science degree in Nuclear Science from the Air Force Institute of Technology in 1982. From there he spent four years in the Technical Operations Division, Air Force Technical Applications Center, McClellan AFB, CA as a nuclear research officer. In 1986, he was selected to obtain his Ph.D. in nuclear engineering, and subsequently entered the program at Texas A&M University. While there, he presented a paper at the 1987 American Nuclear Society convention on comparisons of running the RELAP5 modeling code on various computers. In 1989, he was assigned to the Air Force Technical Applications Center Headquarters at Patrick AFB, FL where he is a branch chief in the Nuclear Technology Directorate over seven other physicists and nuclear engineers. His permanent address is:

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