

Defense Nuclear Agency Alexandria, VA 22310-3398



DNA-TR-89-289

Shock Induced Cavitation

Vincent J. Cushing Cushing Associates, Inc. 153 Williams Drive Annapolis, MD 21401

February 1991

Technical Report

CONTRACT No. DNA 001-80-C-0070

Approved for public release; distribution is unlimited.



91 2 21 009

Destroy this report when it is no longer needed. Do not return to sender.

PLEASE NOTIFY THE DEFENSE NUCLEAR AGENCY, ATTN: CSTI, 5801 TELEGRAPH ROAD, ALEXANDRIA, VA 22310-3398, IF YOUR ADDRESS IS INCORRECT, IF YOU WISH IT DELETED FROM THE DISTRIBUTION LIST, OR IF THE ADDRESSEE IS NO LONGER EMPLOYED BY YOUR ORGANIZATION.



1

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188		
Public reparting burden for this collection of a gathering and mantening the data needed, a collection of information, including suggestion	ind compl 16 for red	eang and revening the selfection of i using this burden, to Weshington Hes	nformation. Sand commony rega dynamore Services. Directorypp fo	dag the bar	ten estimate or any other separat of the Operations and Reports, 1215 Julianess
Done Highway, Suite 1204. Arlington, VA 2220 1. AGENCY USE ONLY (Leave blai	2-4302. 4	nd is the Office of Management and Bu 2. REPORT DATE 910201	and Barrowsk Bash man Barrows	D DATES (COVERED 790106 - 831231
4. TITLE AND SUBTITLE		510201	rechircur		ING NUMBERS
Shock Induced Cavit	atio	n		PE-	NA 001-80-C-0070 62715H H02CAXS
6. AUTHOR(S)	Ŷ		·····	TA-	
Vincent J. Cushing				WU-	DH004592
7. PERFORMING ORGANIZATION	NAME(S) AND ADDRESS(ES)			ORMING ORGANIZATION
Cushing Associates,	, Inc	•			
153 Williams Drive Annapolis, MD 2140	1			CA	101FR
)1				
9. SPONSORING/MONITORING A	GENCY	NAME(S) AND ADDRESS(E	S)		NSORING/MONITORING
Defense Nuclear Age					
6801 Telegraph Road Alexandria, VA 223		398			-TR-89-289
SPWE/Tremba					
11. SUPPLEMENTARY NOTES					
This work was spon B344080462 H02CAXS	sored (388	by the Defense N 01 H25904.	uclear Agency ur	nder RD	T&E RMSS Code
12a. DISTRIBUTION/AVAILABILIT	Y STAT	EMENT	<u> </u>	126. DI	STRIBUTION CODE
Approved for public	: rel	ease; distributi	on is unlimited.		
13. ABSTRACT (Meximum 200 wo	rds)			ļ	
Shock test planning tion as a mechanism	, for	surface ships read damage to naval s	newed interest i nips. To assist	n shoc DNA i	k induced cavita- n formulating and
conducting a progra	am fo	r further investig	gation of cavita	ition,	there is provided
here: a review of a refinement of the	past Porv	theory and exper	imental data, ex er flow in the i	tendin	g back to WWII; t shockwave: a
review of hull plat	te lo	ading involving lo	ocal cavitation;	and	predictions of
secondary pressures tion and sea-bottom	s and 1 ref	impulses generation Sea-bot	ed by combined e tom reflection (ffects	of bulk cavita-
for a specific shock test site are generally known poorly. Analysis indicates that an unfortunate choice of test site could produce large reflection-and-					
cavitation effects.					
14. SUBJECT TERMS				<u></u>	15. NUMBER OF PAGES
Bulk Cavitation	Hu	11 Plate Loading	Spalla	tion	66
Local Cavitation		ock Induced Cavita			16. PRICE CODE
17. SECURITY CLASSIFICATION		CURITY CLASSIFICATION	19. SECURITY CLASSI	CATION	20. LIMITATION OF
OF REPORT UNCLASSIFIED	÷.	F THIS PAGE CLASSIFIED	OF ABSTRACT UNCLASSIFIED		ABSTRACT SAR
NSN 7540-01-280-5500					Standard Form 298 (Rev. 2-89

.

•

Prescribed by ANSI Sta 239-18 298-102

UNCLASSIFIED SECURITY CLASSIFICATION OF THIS PAGE CLASSIFIED BY:

N/A since Unclassified

DECLASSIFY ON: N/A since Unclassified

DISTRIBUTION LIST UPDATE

This mailer is provided to enable DNA to maintain current distribution lists for reports. We would appreciate your providing the requested information.

-	 Add the individual listed to your distribution list. Delete the cited organization/individual. Change of address. 	NOTE: Please return the mailing label from the document so that any additions, changes, corrections or deletions can be made more easily.
ſ	NAME:	
	DRGANIZATION:	
	OLD ADDRESS	CURRENT ADDRESS
	SUBJECT AREA(s) OF INTEREST:	
CUI HERE AND RETURN		
	DNA OR OTHER GOVERNMENT CONTRACT NUMBER: _	
	CERTIFICATION OF NEED-TO-KNOW BY GOVERNMENT SPONSOR (if other than DNA):	
	SPONSORING ORGANIZATION:	
J	CONTRACTING OFFICER OR REPRESENTATIVE:	
	SIGNATURE:	

PREFACE

Shock test planning for the Navy surface ships prompted new interest in the phenomena of Underwater Cavitation as a damage augmenting process. Results have been transmitted via letter reports during the planning process. This is the formal Final Report.

The following tasks were carried out:

- 1. Review past theory and experimental data;
- 2. Extend Bulk Cavitation theory for secondary pressures owing to momentum accretion and afterflow on bulk cavitation region's closure front;
- 3. Extend theory to include hull surface's inertia and compliance;
- 4. Assist in design of bulk cavitation experiments; and monitor DNA's other cavitation efforts.

Predictions are made for possible effects of Bulk Cavitation in a typical shock test.

Technical monitors have been Lt. Robert Elsbernd, USN and Lt. Dennis Sobota, USN.

The Underwater Explosives Research Division of David Taylor Research & Development Center has considerable background information on Bulk Cavitation and Hull Plate Loading. The writer is grateful for recent help provided by that Division's John Wise, Robert Walker and John Gordon, and also for early help by W. W. Murray and Heinrich Schauer.

DZID

Acce	ssion For	
NTIS	GRALI	8
DTIC	TAB	ñ
Unan	nounced	ă
Just	ification.	
	ibution/	Oodea
	Avall and	
Dist	Special	. J
A-1		

CONVERSION TABLE

Conversion factors for U.S. Customary to metric (SI) units of measurement.

MULTIPLY ______ BY _____ TO GET TO GET ______ BY _____ DIVIDE

		·····
angstrom	1.000 000 X E -10	meters (m)
atmosphere (normal)	1.013 25 X E +2	kilo pascal (kPa)
bar	1.000 000 X E +2	kilo pascal (kPa)
barn	1.000 000 X E -28	meter ² (m^2)
British thermal unit	1.054 350 X E +3	joule (J)
(thermochemical)	1.054 550 x 2 + 5	Joure (3)
calorie (thermochemical)	4.184 000	joule (J)
cal (thermochemical)/cm ²	4.184 000 X E -2	mega joule/m ² (MJ/m ²)
curie	3.700 000 X E +1	
		giga becquerel (CBq)*
degree (angle)	1.745 329 X E -2	radian (rad)
degree Fahrenheir	$\tau = (t^{\circ}f + 459.67)/1.8$	degree kelvin (K)
electron volt	1.602 19 X E -19	joule (J)
erg	1.000 000 X E -7	joule (J)
erg/second	1.000 000 X E -7	watt (W)
foot	3.048 000 X E -1	meter (m)
foot-pound-force	1.355 818	joule (J)
gallon (U.S. liquid)	3.785 412 X E -3	meter ³ (m ³)
inch	2.540 000 X E -2	meter (m)
jerk	1.000 000 X E +9	toule (J)
joule/kilogram (J/kg)	1.000 000	Gray (Gy)**
(radiation dose absorbed)		
kilotons	4.183	terajoules
kip (1000 1bf)	4.448 222 X E +3	newton (N)
kip/inch ² (ksi)	6.894 757 X E +3	kilo pascal (kPa)
ktap	$1.000\ 000\ X\ E\ +2$	newton-second/m ²
ktap	1.000 000 x 2 +2	$(N-s/m^2)$
micron	1.000 000 X E -6	meter (m)
mil	$2.540\ 000\ X\ E\ -5$	meter (m)
		meter (m)
mile (international)	1.609 344 X E +3	
ounce	2.834 952 X E -2	kilogram (kg)
pound-force (lbf avoirdupois)	4.448 222	newton (N)
pound-force inch	1.129 848 X E -1	newton-meter (N·m)
pound-force/inch	1.751 268 X E +2	newton/meter (N/m)
pound-force/foot ²	4.788 026 X E -2	kilo pascal (kPa)
<pre>pound-force/inch² (psi)</pre>	6.894 757	kilo pascal (kPa)
pound-mass (1bm_avoirdupois)	4.535 924 X E -1	kilogram (kg)
pound-mass-foot ²	4.214 011 X E -2	kilogram-meter ²
(moment of inertia)	1	(kg•m ²)
pound-mass/foot ³	1.601 846 X E +1	kilogram/meter ³
-		(kg/m^3)
rad (radiation dose absorbed)	1.000 000 X E -2	Gray (Gy)**
roentgen	2.579 760 X E -4	coulomb/kilogram
		(C/kg)
shake	1.000 000 X E -8	second (s)
slug	1.459 390 X E +1	kilogram (kg)
torr (mm Hg, 0°C)	$1.333 22 \times E -1$	kilo pascal (kPa)
corr (und right o o)		

* The becquerel (Bq) is the SI unit of radioactivity; | Bq = | event/s. **The Gray (Gy) is the SI unit of absorbed radiation.

TABLE OF CONTENTS

Sec	tion	Page
	PREFACE	iii
	CONVERSION TABLE	iv
	LIST OF ILLUSTRATIONS	vi
1	INTRODUCTION	1
2	BULK CAVITATION	2
	2.1 Upper Cavitation Depth	2
	2.2 Accretion to Spall	7
	2.3 Lower Cavitation Depth	7
	2.4 Water Hammer	2 2 7 7 8
	2.5 Trapped Energy	8
	2.6 Spherical Explosions	10
3	CAVITATION CLOSURE	14
	3.1 Planar Shockwave	14
	3.2 Spherical Shockwave	15
4	EXPERIMENTAL DATA	22
	4.1 Field Tests	22
	4.2 Laboratory Tests	22
5	BOTTOM REFLECTION	28
	5.1 Theory	28
	5.2 Shock Testing Considerations	33
6	HULL PLATE LOADING	35
	6.1 Observations	35
	6.2 Preliminary Analysis	37
	6.3 Loading at Depth	40
	6.4 Loading near the Surface	40
	6.4.1 Qualitative Description of Plate Motion	45
7	LIST OF REFERENCES	49
	APPENDIX: Bulk Cavitation BASIC Source Listing	53

LIST OF ILLUSTRATIONS

Figure

I

1

1

4

(

1

١

ł

1	Compressive underwater shockwave moving toward free surface	3
2	Total pressure as a function of depth after shockwave reflection	4
3	Total pressure at reflected rarefaction front	5
4	Motion of water particles at various depths	6
5	Approximate water particle trajectories	9
6	Geometry for analysis of spherical explosions	11
7	Variables as a function of horizontal range for spherical explosion	12
8	Variables as a function of horizontal range for spherical explosion	13
9	Spall in neighborhood of radius of first closure	16
10	Mach wave generated by supersonically travelling locus of closure	17
11	Closure mach number and time to closure for spherical explosion	19
12	Water hammer's duration, pressure and impulse for spherical explosion	20
13	Extent of bulk cavitation and locus of closure for spherical explosion	21
14	Spall surface velocity in bulk cavitation test at Mono Lake, 1969	23
15	Laboratory facility simulating nuclear-explosion-like steam bubble	24
16	Laboratory test facility showing schlierin-free observation window	25
17	Sequence of drum camera photos taken in laboratory test facility	27
18	Bottom reflection geometry	29
19	Phase change a and reflectivity for bottom reflection	30
20	Wave distortion for incidence angles beyond the critical	31
21	Friedlander's treatment of spherical pulse reflecting from bottom	32
22	Possible expected pressure history for shock test of ship	34
23	Summary of streak camera tests for shocked circular diaphragm	36
24	(a) Shockwave incident on plate (b) Characteristic times in analysis	38
25	Dimensionless variables as function of time for shocked plate	41
26	Maximum impulse time and reflected wave zero-crossing time	42
27	Impulse as function of time constant ratio r (= t_n/t_n)	44
28	(a) Shock impinging on supported hull plate (b) approximation	46
29	Beginning motion of supported hull plate	48

SHOCK INDUCED CAVITATION

SECTION 1

INTRODUCTION

During shock test planning for the Navy surface ships, there was renewed interest in the phenomenon of bulk cavitation as a damage augmenting process. Results of the subject contract were transmitted via letter reports during the planning process. This is the formal Final Report.

The following tasks were carried out:

- 1. Review past theory and experimental data;
- Extend Bulk Cavitation theory for secondary pressures owing to momentum accretion and afterflow on bulk cavitation region's closure front;
- 3. Extend theory to include hull surface's inertia and compliance;
- 4. Assist in design of bulk cavitation experiments; and monitor DNA's other cavitation efforts.

Predictions are made for possible effects of Bulk Cavitation in a typical shock test.

Bulk Cavitation theory is provided in Section 2. The Closure process, with consequent water hammer, is described in Section 3.

The REFERENCES cite the history of Bulk Cavitation, Local Cavitation and hull plate loading. Past cavitation experiments are discussed in Section 4. Laboratory experiments, using an exploding wire to simulate a nuclear-like steam bubble, are also discussed.

Work to date by all has involved the effects of the direct shock wave only — unfettered by secondary shock waves owing to bottom reflection. Bottom reflections, see Section 5, can play a substantial role. Further investigation is required.

Fundamentals of hull plate loading associated with local cavitation are discussed in Section 6. Quantitative detail involves diffraction analysis, not yet achievable in closed form analysis. Simplified, closed form analyses are discussed. Computer codes (including diffraction phenomena) and results are discussed.

The Appendix lists a Z100 Basic code formulated to determine, for a spherical explosion, the extent of Bulk Cavitation as well as the depth of final closure and intensity of water hammer shocks generated.

SECTION 2

BULK CAVITATION

The phenomena of Bulk Cavitation and Closure have been described in work dating back to 1943¹⁵,²⁶,²⁷,⁴⁵. For a brief description of the phenomena we follow reference 15.

Consider a compressive underwater shockwave moving toward a free water surface as shown in figure 1a. We consider a shock wave with initial pressure p_0 followed by an exponential decay of duration t_w . At distances from the explosive source which are of practical interest, the velocity of the shockwave is essentially the speed of sound, c, in water. Hence, the wavelength λ of the incident shockwave is

$$\lambda = ct_{W} \qquad (1)$$

2.1 UPPER CAVITATION DEPTH.

Just before shock front arrival at the surface, the total absolute pressure is as depicted in figure 1b. When the compressive front arrives at the surface, boundary conditions require that a rarefactive wave be reflected. Just after reflection from the surface, the total pressure is as depicted in figure 2. We note that as the rarefactive front, of strength $-p_0$, moves down into the water: (1) the pressure beneath the rarefaction front is that owing to the waning portions of the incoming wave; (2) above the rarefaction front, the total pressure remains p_a (atmospheric pressure) at the surface and progressively falls as the rarefaction front is approached. Below some point, shown as Z_0 , the total absolute pressure is negative — the water goes into tension.

Since seawater cannot withstand appreciable tension, the water ruptures; the pressure in the ruptured water falls to vapor pressure — zero for all practical purposes.

All types of incident shock waves do not necessarily result in cavitation. If the peak pressure is sufficiently large and the decay wavelength λ is exceedingly long, hydrostatic pressure will rise faster than the incident wave decays, and total pressure will monotonically increase with depth — as indicated by figure 3's curve III. A shorter wavelength may result in pressure lowering as shown in curve II, but zero pressure is never reached. Most practicable explosions, including nuclear, have pressure/wavelength combinations (by the time the free water surface is encountered) such that zero pressure is achieved and bulk cavitation commences at some depth Z_0 .

Figure 4 depicts an incident shockwave rising vertically to the free water surface Z_s , which is initially at zero depth. If the incident shock has a particle velocity of u_0 , upon reflection the surface Z_s initially moves upward with velocity $2u_0$. When the reflected wave reaches the onset depth Z_0 , the water ruptures; total pressure at the depth is substantially zero — the vapor pressure of water at ambient temperature.



Figure 1. (a) Compressive underwater shockwave moving toward a free surface. Peak pressure p_0 ; decay time constant t_w . (b) Total absolute pressure just before shock arrival at free surface.





Figure 3. Total pressure at reflected rarefaction front as a function of depth z.



Figure 4. Motion of water particles at various depths, ranging from z_s (the water surface) to z_0 (the depth of cavitation onset), z_c (depth of ultimate cavitation closure) and z_T (terminal depth beyond which there is no cavitation). The particle originally at depth z_1 is typical of the water particle which is accreted to the lower boundary z_L of the spall. The water particle of depth z_2 follows a ballistic trajectory till it falls back to its original depth.

2.2 ACCRETION TO SPALL.

At that instant there is a liquid layer of water — a spall of thickness (Z_s-Z_0) — moving upward with almost the velocity $2u_0$. The decelerating forces are two: (1) boundary forces — atmospheric pressure at the upper surface and negligible vapor pressure at the lower surface; and (2) body forces — gravitational force acting throughout the liquid layer.

As shown in figure 4, the spall's lower surface is denoted by Z_L . (Z_S-Z_L) — the spall's thickness — is initially equal to (Z_S-Z_O) . We shall see that this thickness grows. As it does the added deceleration owing to atmospheric pressure at the upper surface becomes relatively less compared with the body force of gravity. Hence the spall deceleration will asymptotically approach 1g.

The particle velocity u at any point of the incoming shockwave is, since the acoustic approximation is adequate

$$\mathbf{u} = \mathbf{p}/\mathbf{\rho}\mathbf{c} \quad , \tag{2}$$

where ρ is the water density, and c is the speed of sound. As the reflected rarefaction front moves into water deeper than that at Z₀, it encounters the waning portions of the incoming wave — where the particle velocity has fallen to u, with u described by equation 2. Accordingly, the rarefaction front leaves behind it cavitated water whose initial upward velocity, or launch velocity, is 2u. Hence the launch velocity continuously falls as the rarefaction penetrates the waning portions of the incoming wave.

Since the cavitated region is entirely at vapor pressure, the deceleration of the launched, cavitated water is not affected by pressure forces, only gravitational forces. Hence each cavitated water particle follows a ballistic trajectory — with exactly one g of deceleration.

We recall that the liquid spall decelerates asymptotically toward 1g - at all times it decelerates faster than 1g. Hence the liquid spall is continuously decelerating into underlying cavitated water. The spall thickness grows owing to accretion of the underlying cavitated water. Figure 4 depicts that the spall's lower surface Z_L is continuously colliding with underlying spalled water. Spall growth continues in this fashion until it accretes the cavitated particle which had been launched from depth Z_c , from what we shall see is the cavitation closure depth.

2.3 LOWER CAVITATION DEPTH.

The reflected rarefaction front, as shown in figure 4, continually produces cavitation until some lower depth Z_T , the cavitation termination depth. Beyond that depth the hydrostatic pressure is too large to allow the total absolute pressure to reach zero. The water below Z_T remains liquid at all times.

The cavitated water just above Z_T has a very small launch velocity and a very brief ballistic trajectory. It soon falls back on the underlying liquid water. Analogous to

behavior of the accreting spall, the underlying liquid surface moves upward wing to cumulative accretion of water that had been in ballistic trajectory in the cavitated region.

2.4 WATER HAMMER,

The lower cavitation boundary rises while the spall's lower surface Z_L falls — until eventually there is no cavitated region left. The particle velocity on the lower cavitation boundary is, to a first approximation, zero. The particle velocity on the upper cavitation boundary is effectively the velocity of the decelerating spall. Hence at time t_c there is a water hammer as the spall impacts the quiescent lower boundary. (Actually the particle velocity of the lower boundary is not exactly zero. The computer program listed in the Appendix takes the actual velocity into account.)

Figure 5 is identical to figure 4 — except that the velocity of the incoming and reflected fronts is assumed to be infinite; i.e., the duration of the cavitationproducing process is assumed negligible compared with spall flight duration and the cavitation closure and water hammer processes. This simplification has enabled a closed form solution 15,46 of the bulk cavitation process. Results compare well with experiment 43,47 .

As shown in figure 4, time t is measured from the moment the incoming shockwave reaches the water surface. Depth Z is measured from the initial position of the free surface.

The figure shows the Z-t trajectories of the incoming shock front and also the reflected rarefaction front. These waves move at velocity c, the speed of sound in water about 1,500 m/sec. Cavitation conditions exist when the reflected wave reaches the onset depth Z_0 . Below that depth the water vaporizes when the rarefaction front passes.

Ultimately the upper, spalled layer will hammer the underlying liquid water beneath the boundary Z_B ; the dynamics of the process produce secondary underwater pressure waves owing to this progressive water hammer.

2.5 TRAPPED ENERGY.

Depending on explosive yield and depth of burst, a large fraction of the total incident shock wave energy can be stored temporarily as kinetic energy and/or gravitational potential energy (of the overlying spall and also of the cavitated particles) during the bulk cavitation process; and the stored energy is re-emitted later during the water hammer process when the cavitation closes.

Bulk Cavitation's late-time secondary waves have produced extensive damage to ships during tests at HARDTACK Umbrella and in subsequent HE tests. This prompted earlier investigations. The program for the ARKANSAS shock tests prompted DNA's renewed interest in Bulk Cavitation.



Figure 5. Water particle trajectories under the approximation that the transit time of the rupturing rarefaction front is negligible compared with the trajectory times of the cavitated water.

2.6 SPHERICAL EXPLOSIONS.

Figure 6 shows the more practical situation of a spherical explosion. The incident shock first arrives at the surface directly above the explosive, at surface-zero (SZ). Launch velocity of the spall is maximum there: it takes maximum time for the spall to complete its trajectory before impacting with the underlying water.

As we increase horizontal distance from SZ, the incident shock arrives later, it is weaker and its angle of incidence with the surface results in a decreasing spall launch velocity. The consequence is in figure 7, which shows the time of closure T_c as a function of horizontal radius find surface-zero. Also shown is the time of arrival T_1 of the incident shockwave. We see that cavitation closure first occurs at radius X_{c1} at time T_{c1} .

Figure 8 shows the maximum surface excursion and the time of maximum excursion for this explosion configuration.



Figure 6. Geometry for analysis of spherical explosions.



Figure 7. Some variables as a function of horizontal range for a spherical explosion.





SECTION 3

CAVITATION CLOSURE

3.1 PLANAR SHOCKWAVE.

An underwater shockwave, at distances from the explosive source which are of interest, behaves virtually as an acoustic wave. In practice the wave has a shock rise to peak pressure p_0 , followed by an exponential-like decay. When such an acoustic wave is reflected from a rigid surface (1) the wave's momentum (total impulse) is reversed, and (2) the wave's kinetic energy is conserved. A corollary of this is that the wave's shape (magnitude and duration) is preserved.

When an underwater shock wave is reflected from a free surface, where negative pressure cannot exist in the medium, spallation or cavitation occurs. The wave's energy becomes stored in the gravitational energy of the spall, and the energy is returned (with considerable dissipation) during the closure process. The result is, eventually, a reflected wave which is compressive. The wave's (1) kinetic energy is not conserved, and (2) while the magnitude of momentum is conserved, the momentum is reversed. A corollary of this is that wave shape is not preserved.

For analysis we assume the Instant Cavitation depicted in Section 2's figure 5 (page 9); i.e., the travel time of the cavitating wave front — which moves at sound speed — is negligible compared with the duration of spallation and cavitation. The particle velocity u(Z) at depth Z in the cavitated region (for incident wave with shock rise to pressure p_0 , and exponential decay with wavelength λ) is ¹⁵

$$u(Z) = -(p_a + \rho g Z + 2p_0 e^{-2Z/\lambda})/\rho c , \qquad (3)$$

where p_a is atmospheric pressure, $\rho g Z$ is hydrostatic pressure at depth Z and ρc is the water's acoustic impedance.

In the cavitated region the pressure is constant vapor pressure of water. There is no pressure gradient; hence, each cavitated water particle follows a ballistic trajectory. The particle's flight time (the time until gravity reverses the particle's momentum and it falls back to its original depth Z) is

$$t_f = 2u(Z)/g$$
 . (4)

For water at depths above the closure point, the water particle doesn't get a chance to complete its ballistic flight; the higher acceleration (owing to the added force of atmospheric pressure on top) of the overlying spall causes the particle to accrete to the spall's underside before flight time is completed. Particles launched at the closure depth complete the ballistic flight. Particles launched below the closure point also do not complete the ballistic flightµ they accrete to the rising underlying water surface. The front of this accretion-by-fallback is the bulk cavitation lower closure front.

The water particle at closure depth Z_c is the last particle to complete its bal-

listic flight time. That particle accretes its mass and momentum at time

$$t_f(Z_c) = 2u(Z_c)/g$$
 (5)

For short wavelength shock waves, the hydrostatic pressure and the atmospheric pressure are negligible compared with the 'pressure' component (see equation 2). Hence the flight time is

$$t_f(Z_c) = 4p_0/(\rho g c)e^{-2Z_c/\lambda}$$
(6)

3.2 SPHERICAL SHOCKWAVE.

In Section 2 it was pointed out that there is a radius (about surface zero) where cavitation closure and water hammer first occur. Figure 9 shows the spall and the underlying water in the neighborhood of this radius X_{c1} of first closure. Figure 9a depicts the situation just prior to first impact; 9b depicts it immediately after first impact. Closure progresses inward with instantaneous contact point X_i , and outward with contact point X_0 .

Immediately after first impact, X_i and X_o travel approximately horizontally with infinite velocity and thereafter progressively slow down.

The outward travelling point X_0 cannot move slower than the horizontal velocity of the incident shockwave, which travels at the speed of sound in water. Thus the outward travelling contact point monotonically slows to mach 1. However, as it approaches mach 1 the distance from the explosion center and the angle of incidence with the surface are such that the energy stored in the cavitation process is negligible, and the consequent water hammer at closure is negligible.

The inward travelling closure point X_i is not limited by the speed of sound in water. It may fall to and below mach 1. Distance and angle of incidence considerations permit appreciable stored energy and final water hammer. Indeed, the horizontal radius X_{m1} at which the contact point is travelling at mach 1 is particularly hazardous since the hammer pressure theoretically increases without bound at that radius¹⁵.

In simple analyses it is assumed that the horizontal motion of cavitated water particles is negligible compared with the vertical motion. Figure 10 depicts the mach wave generated by the supersonically (in water) travelling closure point. Figure 10 shows the vertical velocities of the spall (u_s) and the underlying water (u_u) . Figure 10's equation 4 shows the particle velocity normal to the mach front and to the reflected mach front. Equation 6 shows that the hammer pressure P_h increases without bound as the closure mach number M slows down and approaches unity. Even so, equations 9 and 10 show that the hammer impulse I_o below closure depth Z_c is bounded. Above closure depth the hammer impulse I is proportional to the depth Z below the water surface.

If a vessel's structure is pressure sensitive, it is at risk of self damage if it launches an underwater depth charge so as to place itself at the mach 1 distance. A



Figure 9. Spall and underlying water in the neighborhood of radius of first closure, X_{c1}, where water hammer first occurs.





target vessel is most vulnerable if placed at that location. There is less hazard if the vessel's structure is impulse sensitive.

Figure 11 graphs the closure mach number and the time of closure for a 40,000 LB. HBX charge. The right ordinate is negative for the inward travelling mach front, and positive for the outward front.

Figure 12 graphs the water hammer's duration, pressure and impulse for the same explosion configuration.

Figure 13 shows the extent of bulk cavitation and also the locus of closure. The data for figure 13 resulted from the computer code given in the Appendix. The Appendix' code is an embellishment of the fortran code developed in reference 4. As shown in the Appendix, the code was modified in reference 9 to account for bottom layer acceleration, and modified for the present report to account for afterflow (10).



Closure mach number and time of closure for spherical explosion. Figure 11.





•





21

ł

SECTION 4

EXPERIMENTAL DATA

4.1 FIELD TESTS.

There have been three field tests specifically designed to investigate bulk cavitation. The first was conducted at UERD in 1962 and reported by Walker and Gorden (ref 47). Theoretical interpretation of its data has been good — until arrival of the first bottom reflections. Thereafter data seems to be the superposition of confusing reverberations.

The second set consisted of the NOL Tests at Mono Lake conducted in 1969 (ref 43). Again, they produced data agreeing reasonably with theory during early times. The Mono Lake tests placed floats with a seismic-suspension integrating accelerometer to measure surface velocity. Figure 14 shows typical results. The integrating accelerometer displayed excessive integration drift about 70 msec after onset of motion.

Data from the Mono Lake tests must be used judiciously owing to unusual sound propagation behavior associated with 1) high salt concentration; 2) substantial concentration of brine shrimp.

The third and more recent tests were conducted by NRL and UERD. They included a more recently developed capability to place PV gages under water — providing useful correlations between pressure and particle velocity throughout the field.

Bulk Cavitation theories do not accommodate bottom reflection (discussed in Section 5). If there is to be a full understanding of Bulk Cavitation it will be necessary to have data from at least one test in deep water, where measurements are not confounded early on by bottom reflections. Enough depth is needed to make sure that a complete cycle of bulk cavitation — incident pressure, generation of cavitation, closure, and measurement of secondary waves — is completed before arrival of the bottom reflection.

4.2 LABORATORY TESTS.

Laboratory tests were carried out in 1969 using an exploding wire as the explosive source ³⁹. HE tests produce a gas bubble consisting of HE combustion products, whereas the exploding wire is nuclear-like in that it generates a steam bubble.

The facility's original use was to gather information on bulk cavitation. It was later put to use to study (1) shockwave/hull-plate interaction with and without local cavitation, and (2) surface wave generation.

The facility is shown in figure 15. The test vessel is a steel cylinder 6 feet in diameter, with near hemispherical ends to permit vacuum or pressurization to 30 psig. The interior is lined with anechoic material.

The principal observation windows are diametrically opposite, 12 inches in diameter







Figure 15. The rationary beach test facility usual exploding wire to simulate nuclear explosion steam bubble.

and consist of schlierin-free glass. One such window is shown in figure 16.

Depending on application, either schlierin or shadowgraph photos are taken. Light sources available are (a) a 1 μ sec spark, (b) a 1 msec xenon flashlamp, and (c) a steady high pressure mercury arc lamp.

The cameras for schlierin work are a single frame 4 X 5 camera as well as a drum camera taking 224 images with frame separation of 40 μ sec and shutter speed of 1 μ sec. Figure 17 shows a typical sequence of drum camera photos.

For longer sequence durations a 35 mm Fastax camera was used — with frame rates up to 3000/sec. A 16 mm Bolex was also available for very long events.

Miniature piezoelectric hydrophones are used to gather simultaneous pressure-time information.

Energy for the explosion was provided by a 20-KV, $15-\mu F$, 5-nH capacitor. The exploding wire can be positioned anywhere in the tank — above or below the water surface. A 2 mm length of 40AWG nichrome was typically used — virtually a point source for most applications.

The 'nuclear' yield of the exploding wire was of the order of 0.04 gm of TNT. Data analysis showed scaling laws applied favorably over a very wide range — for full scale nuclear (CROSSROADS-Baker, at the same scaled depth of burst in shallow water), large HE (to 104 lbs of TNT) and small HE charges (to 10^{-4} lbs of TNT).²⁸

With the exploding wire as a simulator for nuclear explosions, the schlierin optics show the progress of direct and reflected shock waves in the water and the air and permit observation of bottom effects and surface effects. Data connecting the generation of surface waves, plumes and air blast have been obtained at low cost. A variety of reflecting bottoms — in material and contour — can be studied.



Figure 16. Exploding wire test facility showing one of the schlierin-free observation windows.



Figure 17. Sequence of drum camera photos taken in exploding wire test facility.
SECTION 5

BOTTOM REFLECTION

5.1 THEORY.

Officer³⁵ is a good source of information on acoustic bottom reflection. Figures 18 through 21 are interesting excerpts. Figure 18 shows the geometry for explosive source, bottom and receiver. ρ is the density of the medium and c is the speed of sound. For incidence angles less than the critical, θ_c , the reflection coefficient is less than 100 percent, i.e., there is some energy transmitted into the bottom. Beyond the critical angle there is total reflection.

Fig. 19 shows an example of the reflectivity coefficient and phase change a as a function of incidence angle θ — for sinusoidal sound trains. The curves are typical for the usual bottom condition where $c_2 > c_1$ and $\rho_2 c_2 > \rho_1 c_1$ — where c_1 and c_2 are the sound speeds in water and in the bottom respectively, and $\rho_1 c_1$ and $\rho_2 c_2$ are the respective acoustic impedances in these media. In such cases reflectivity reaches 100 percent at some critical incidence angle θ_c ; for greater angles the reflectivity remains at 100 percent while the phase α of the wave train begins to shift — reaching 180° phase shift at grazing incidence, $\theta = 90^\circ$.

Such phase shifting causes distortion of an incident wave. For the practical case of a wave with a shock rise followed by an exponential-like decay, reference 2 describes the distortion for angles of incidence beyond the critical, as shown in figure 20. (N.B, the angle α shown in figure 20 is the phase change angle, which is related to the angle of incidence Θ as typically shown in figure 19.) The phase angle remains zero — and the wave is undistorted — from normal incidence all the way to the critical incidence angle. Thereafter the phase begins to shift and the wave distortion is as shown in figure 20.

When the phase shift is larger than about 90° for this explosion-like wave shape, one observes a compressive precursor (for times t < 0), and a rarefaction follow-up — and these are for for bottom incidence angles generally encountered in underwater explosion tests. This may explain the 'strange' bottom reflection rarefactions noted in Walker and Gordon's 1962 tests.

Friedlander¹⁹ provides theoretical treatment of explosion-like pulses. Figure 21 shows his treatment of a spherical pulse reflecting from a bottom. Reflection front B is for less than critical angle; front L is at the critical angle. Front LEN is beyond the critical angle. For example, along the line ED, we have (1) a gradual rise compressive precursor (compare Figure 20) between E and the 'reflected front' and (2) between the reflected front and D there is a main wave, compressive or rarefactive (compare Fig. 20). LEN is an unequivocal precursor for that portion beyond the incident, near N.



Figure 18. Bottom reflection geometry, showing explosive source, bottom and receiver.



Figure 19. Phase change a and reflectivity coefficient as a function of incidence angle Θ — for sinusoidal sound trains.









ł

5.2 SHOCK TESTING CONSIDERATIONS.

Bottom reflections can play a significant role in shock testing. Lacking adequate knowledge of bottom conditions, one must meantime consider what can happen with 100 percent reflection, coupled with bulk cavitation effects.

Among the test configurations	considered was the following:
Explosive weight:	40,000 lbs HBX
Explosive depth:	200 feet
Water depth:	700 feet
Horizontal Range:	722 feet

Using the analysis of reference 15 (embodied in the program listed in the Appendix), at a horizontal range of 722 feet the bottom reflected wave would appear to cause a second round of bulk cavitation (beginning just after closure of the first cavitation). The impact velocity — and consequent closure hammer pressure — is larger than that of the first cavitation closure. Further, the geometry is such that the second cavitation closure first occurs at a horizontal range of 1800 feet; however, the mach 1 closure point where peak pressure increases without bound — occurs at a horizontal range of 720 feet.

At a horizontal range of 722 feet, the surface incidence angle of the main shock is about 75°. The surface incidence angle of the bottom reflection is about 30° . Even though the reflected pressure is only half (assuming unity bottom reflectivity) of the main pressure, the change in incident angle more than compensates, generating higher spall velocity and consequently higher closure hammer velocity.

Figure 22 shows the expected pressure history at a depth of 21 feet and horizontal range of 722 feet. The second cavitation closure pressure is theoretically infinite; however, the impulse is finite, estimated to be about 3 psi-seconds (compared with a value of 0.92 for the impulse owing to closure of the first cavitation).

Without adequate knowledge of bottom conditions in proposed shock testing it is not possible to predict whether or not there might be a surprisingly large impulse delivered owing to interaction of bottom reflection and bulk cavitation.





SECTION 6

HULL PLATE LOADING

A prime concern for underwater shockwave/ship interaction is the effects on equipment within a ship. Secondly, observed hull plate buckling is enough to cause flow noise that compromises a ship's sonar ability. The mechanisms of hull plate buckling are not adequately understood.

Analytical solutions to the problem were first sought during WWII.²⁵,²⁷,³¹,⁴⁵ Tests and analysis pointed to local cavitation playing a major role. The underwater Explosives Research Division of NSRDC conducted a number of HE tests during the 1960s to study the mechanisms of hull damage from underwater explosions. At that time it was realized that there were three intertwined phenomena contributing to hull plate damage: (1) initial shockwave/hull-plate interaction, (2) local (and perhaps also bulk) cavitation and closure, and (3) diffraction.

The complexity requires machine numerical methods. Even so, a semi-quantitative analysis — not involving diffraction — enabled interpretation of the UERD data.¹²

Though no analytical solution is available, the differential equations representing the problem can be set up and solved digitally. Useful results have been obtained by a number of workers for the situation where the water is always an acoustic medium, i.e, at depth where ambient pressure allows large negative acoustic pressures.⁴,⁶,¹⁷,²¹,⁻ $^{22},^{23},^{29},^{36}$

More recent numerical treatment includes provision for a bilinear acoustic medium interacting with simplified structural configurations.⁵,¹⁸,³³,³⁴,³⁷

Original insights into hull plate loading stylized the interaction to be that of shockwave and plate.⁴⁵ Late time effects in these analyses pointed toward local cavitation as a major affect²⁵. Afterflow was also considered as a late time loading mechanism.⁴¹

6.1 OBSERVATIONS

Streak camera tests have been conducted on a circular diaphragm — held on a test hull bottom by a heavy rim — and subjected to an HE shockwave.²⁴ The results are summarily sketched in figure 23. The first sketch shows the diaphragm immediately after it has been struck by the shock front — it has just begun to move. The space between the diaphragm and the reflected shock front has been expanded in the adjacent box which shows the pressure distribution. There, we see that the reflected shock front is moving into the waning portions of the initial, incoming shockwave. The sum of pressures in the incident wave and the reflected wave cause substantially a pressure doubling.

As the diaphragm accelerates to higher velocity, the reflected wave, initially a compression, becomes increasingly rarefactive. In the pressure distribution box, we therefore observe the rapidly decreasing total pressure. If water could withstand tension, the total pressure would follow the dotted line. However, it cannot; hence, near the



Figure 23. Summary of streak camera tests showing response of circular diaphragm to incident shockwave.

diaphragm the water cavitates — by forming vapor bubbles, or by forming water droplets in a matrix of water vapor. Whichever way, the pressure in the cavitated region is that of water vapor, effectively zero for our purposes.

Sketch 2 shows the diaphragm at a later time. The flat central portion of the diaphragm does not yet know that the edges have been clamped by the heavy rim; a bending wave, travelling inward, carries this information. The streak camera record displays a jerky motion as shown in the second and later sketches. Theoretical work has not shown a cause for this jerkiness. It is conjectured that a water layer, still attached to the diaphragm, is vibrating.

Later sketches show further motion and progress of the bending wave. Next to last sketch shows the bending wave reaching the center. The camera record indicates that all motion has momentarily stopped.

Then, at a substantially later time, the diaphragm is bulged outward. It appears that this late impulse is a water hammer that takes place at cavitation closure.

6.2 PRELIMINARY ANALYSIS.

Kennard has displayed the physics of shockwave/hull interaction involved quite well by a variety of analytical treatments with restricting assumptions.²⁷

Four characteristic times are used (see figure 24b):

- 1. t_w wave time the incident wave is assumed for analytical purposes to have a shock rise to pressure po followed by exponential decay with time constant t_w.
- 2. t_m maximum time required for the hull structure being analyzed to reach maximum velocity.
- 3. t_d diffraction time the time required for an underwater acoustic wave to propagate from the structure's center to its edge.
- 4. t_s swing time the time required for the structure to reach maximum deflection and come to rest.

The simplest analysis showing hull/cavitation phenomena is that of an exponentiallike shock wave striking a plate of infinite lateral extent, as shown in figure 24a.^{27,45}

If the areal density of the plate is m, a damping time constant t_p comes into play, defined by

$$t_{p} = m/\rho c \qquad (7)$$

 t_p may be visualized as the time required for a sound wave to traverse a thickness of water having the same mass as the plate.²⁷ For example, a one-inch steel plate has a damping time of 0.129 msec; a rigid hull drawing 30 feet of water has an effective damping time of 6 msec (for a wave shocking it from the bottom).

In figure 24a we see the incident shock wave. At any instant at the plate the incident plus reflected shock wave sum to a pressure doubling, i.e, a pressure of $2p_i$.



Figure 24. (a) Exponential-like shockwave incident on plate of infinite lateral extent. (b) Characteristic times employed in analysis of shocked plate.

Superposed on this is a rarefaction wave owing to motion of the plate. As indicated in figure 24a, the magnitude of the rarefaction is $\rho c \cdot dz/dt$ — where ρ is the water density (65 lbs/ft³ or 2 slugs/ft³), c is the speed of sound in water (5000 fps), and z is the displacement of the plate. ρc is the acoustic impedance of water.

The equation of motion for the problem shown in figure 24a is

$$\mathbf{m} \cdot \mathbf{d}^2 \mathbf{z} / \mathbf{d} \mathbf{t}^2 = 2\mathbf{p}_i - \rho \mathbf{c} \cdot \mathbf{d} \mathbf{z} / \mathbf{d} \mathbf{t} \quad . \tag{8}$$

The incident shock pressure is described by

$$p_i = p_0 e^{-t/t_w}$$

Wave time t_w and peak pressure p_0 follow empirical scaling laws¹⁰:

$$t_{w} = MW^{1/3} (R/W^{1/3})^{m} , \qquad (10)$$

$$p_0 = K(W^{1/3}/R)^k$$
, (11)

where R is the radius from the explosive and the empirical constants are shown in Table 1 for various explosives.¹⁰,¹¹,⁴³

Table 1				
Explosive	K, psi	k	M, msec	m
HBX-1 Lithanol Pentolite INT	23,980 13,180 22,500 21,600	1.13 1.13 1.13 1.13	0.0654 0.12 0.0969 0.0676	0.18 0.18 0.08 0.24

The ensuing graphs of pressure and motion of the shockwave/plate interaction are plotted in terms of the following dimensionless variables:

$r = t_p / t_w$,	•	(12a)
P "		

 $P_i = p_i/p_0$, (12b)

 $P_t = p_t/p_0 , \qquad (12c)$

 $\mathbf{P_r} = \mathbf{P_t} - \mathbf{P_r} \quad , \tag{12d}$

$$U = \frac{1}{2}/(p_0/\rho c)$$
, (plate velocity) (12e)

$$T = t/t_{W} , \qquad (12f)$$

$$KE = \int_{0}^{1} P_{t} \cdot U \cdot dT , \qquad (kinetic energy of plate) \qquad (12g)$$
$$Z = z/\lambda , \qquad (12h)$$

where shock wavelength λ is related to t_w by

$$\lambda = c \cdot t_{\mathbf{W}} , \qquad (12i)$$

and c is the speed of sound in water.

In terms of these the dimensionless input shockwave is expressed by

$$P_i = e^{-T} , \qquad (13a)$$

and the various responses are:

$$P_{t} = 2[e^{-T/r} - re^{-T}]/(1-r) , \qquad (13b)$$

$$\mathbf{P}_{\mathbf{r}} = \mathbf{P}_{\mathbf{t}} - \mathbf{P}_{\mathbf{r}} \quad , \tag{13c}$$

$$U = 2[e^{-T} - e^{-T/r}]/(1-r)$$
 (13d)

These are plotted in figure 25 for respective r values of 0, 1/4, 1 and 4.

6.3 LOADING AT DEPTH.

If plate loading takes place at sufficient depth — ambient pressure (atmospheric plus hydrostatic) is large — then the acoustic wave negative pressures are realizeable: the entire plate loading process can be described in terms of the foregoing graphs.

The area under the P_t curve is the impulse imparted to the surface plate. The impulse is maximum at some time T_{to} , depending on the value of the parameter r. Thereafter P_t goes negative and thereby begins decelerating the surface plate; it begins to withdraw impulse from the plate. Indeed if we wait several time constants — let T approach infinity — the acoustic pressures will withdraw all impulse loading from the plate. The plate will come to rest.

Figure 26 shows values of T_{tc} for values of r.

 P_t , the reflected pressure, overall has a total (dimensionless) impulse of zero. Hence, the total impulse of P_r is opposite to that of P_i . At depth, then, the reflected pressure can radiate downward to infinity, carrying away energy and momentum, ultimately leaving the plate motionless.

Hence in a totally acoustic plate loading, the maximum impulse (up to a maximum of -2 for large r) is imparted to the plate at time T_{to} . Thereafter the acoustic waves proceed to unload the plate, ultimately: (1) leaving it with zero imparted impulse; (2) delivering all the incident wave's impulse to the reflected wave's impulse, which radiates away to infinity.

6.4 LOADING NEAR THE SURFACE.

If we are not at significant depth — i.e., a vessel at or near the surface — the above described acoustic waves can be followed only up until approximately T_{to} , when the total pressure passes through zero. Thereafter the total acoustic pressure P_t cannot continue into its negative phase; the water rv_t zures and cavitation takes place.







 $I_{t} = \int_{1}^{1} P_{t} \cdot dT$





We can use these acoustic curves to find the condition of the plate at time T_{to} , when the plate has received maximum impulse. After cavitation the actual pressure P_t remains at zero (actually the vapor pressure of water — practically zero); no impulse is subsequently withdrawn from the plate.

By time T_0 the reflected wave P_r has become negative. This reflected wave is travelling down into the water; hence, its argument as a travelling wave is T-Z (dimensionless time minus dimensionless displacement). Since this reflected pressure wave $P_r(T-Z)$ is propagating back down into the water, it moves into the waning portions of the incoming $P_i(T + Z)$. Thus at some place and time, $P_r(T-Z)$ plus $P_i(T+Z)$ becomes negative -- physically not allowed at shallow depths (where ambient pressure -- sum of atmospheric and hydrostatic pressures -- is effectively zero); the water cavitates. As time progresses the boundaries of this growing cavitated region are determined by the two values of Z that approximately satisfy, at each time $T^{11,15,46}$

 $P_i(T+Z) + P_r(T-Z) = 0$ (14)

The only portion of $P_r(T-Z)$ that ultimately escapes the cavitation process is the compressive front, which is seen in figure 25 to end at some time T_{ro} , depending on the value of the parameter r.

Beyond T_{ro} figure 25 shows that the additional impulse in the reflected wave $P_r(T-Z)$ is negative. All of this negative impulse — initially the tail of $P_r(T-Z)$ radiating down away from the plate — cannot escape and ultimately is deposited in a spall of cavitated water beneath the plate.

There are four impulses to consider when the loading and cavitation processes have been completed (but prior to the beginning of the cavitation closure process, i.e., the cavitation closure phase of further loading of the surface plate):

I_i - the total impulse of the incident compression wave (equal to unity in dimensionless terms);
 I_{to} - the total impulse imparted to the surface plate by the reflected wave;
 I_{ro} - the total impulse (which escapes to infinity) in the compressive front of the reflected wave; and
 I_s - the total impulse or momentum in the cavitated slug of water immediately beneath the surface plate.

The mathematical definitions of these are shown in figure 27. Momentum conservation requires

$$I_i + I_{to} + I_{ro} + I_s = 0$$
 (15)

These impulses, as a function of the time constant ratio r, are graphed in Figure 27. Also shown, in Figure 26, are the pertinent times, T_{to} and T_{ro} , as a function of r.

r is small if (1) the surface load m is sufficiently small (including zero, the situation for Bulk Cavitation), regardless of t_w (i.e., for small or large explosive charges); or, (2) t_w is large (corresponding to large scale HE and nuclear).

For large r cavitation plays no important role - negligible incident impulse is deposited in a cavitated spall - cavitation is effectively nonexistent. Behavior is





similar to the well known results for acoustic reflection from a rigid wall: (1) the reflected wave has an impulse equal in magnitude, opposite in direction, to the incident impulse; and (2) the wall receives an impulse equal to twice the incident impulse.

6.4.1 QUALITATIVE DESCRIPTION OF PLATE MOTION.

Figure 28a shows a shock wave impinging perpendicular to hull plating supported by stringers. As an approximation we consider the configuration of figure 28b. Instead of a diaphragm with bending wave, we connect the light weight hull plating by means of a spring to the heavy, rigid hull frame. We look at the beginning motion, while the effect of the spring is small.

Figure 28b implies that the fraction f of the water/hull interface surface has surface load m; fraction (1-f) has surface load M. The simplification we employ is that the total acoustic pressure p_t is at each time the same in (1) the area fraction f, and (2) the area fraction (1-f). In other words we assume that it takes negligible time for pressures to diffract between the area fraction f and the area fraction (1-f).

An example that was considered consists of a 40,000 pound charge of HBX-1 at a distance of 600 to 800 feet from a ship with 1/2 inch plating and drawing 20 feet of water. Then we have

P _O	500	psi
tw	2.9	msec
ρD	1,280	lbs/ft ²
f	0.8	
m	20.8	lbs/ft ²

The overall ship mass is ρD where ρ is the density of seawater and D is the ship draft. The loadings of the partial areas f and (1-f) must collectively support this mass:

$$f \cdot m + (1-f) \cdot M = \rho D$$
, (16)

so that

$$M = (\rho D - f \cdot m) / (1 - f) .$$
(17)

Since the hull plate mass m is generally very small compared with the ship mass, we have to a good approximation

$$M = \rho D / (1-f)$$
 (18)

If the unsupported portion of the hull plating effectively makes up 80 percent of the hull area, then the massive loading M is 5 times the ship mass per unit area ρD :

M 6,400 1bs/ft2.



Figure 28. (a) Shock impinging on supported hull plate. (b) Analytical approximation of hull plate.

The damping time constants for m and M are

t _{pm}	2.08	msec
t _{DM}	640	msec

and the dimensionless ratios r are

rm	0.72	
r _M	220	

With these values we see from the graphs of figures 25 through 27 that the stiff hull portions are initially absorbing all the initial impulse; the reflected wave is high pressure. The compliant plate portion rapidly begins to radiate a low pressure wave and traps much of its energy in the kinetic energy of spalled or cavitated water.

As seen in figure 29, the high pressure reflected wave pumps its energy into the low pressure region, exacerbating the trapped kinetic energy there. At some late time this kinetic energy must be absorbed by the hull plate.

Descriptions of major effects have been provided here to the extent feasible without explicitly including diffraction time considerations in the analysis. Rigorous answers require machine numerical solution.







SECTION 7

LIST OF REFERENCES

- Anon, "The Underwater Shock Analysis (USA) Code, A Reference Manual," Lockheed Palo Alto Research Laboratory Final Report DNA 4524F to Defense Nuclear Agency, 28 Feb. 1978.
- 2. Arons, A. B. and D. R. Yennie, JASA, 22, 1950.
- 3. Friedlander, F. G, SOUND PULSES, Cambridge University Press, New York 1958.
- 4. Bedrosian, B, and F. L. DiMaggio, "Acoustic Approximations in Fluid-Shell Interactions," J. Engr. Mech. Div. ASCE, 731, June, 1972.
- 5. Bleich, H. H, and I. S. Sandler, "Interaction between Structures and Bilinear Fluids," Int. J. Solids and Structures, 66, 617, 1970.
- 6. Chertock, G, "Transient Flexural Vibrations of Ship-Like Structures Exposed to Underwater Explosions," JASA, 48 (1) (Part 2), 170, 1970.
- 7. Clarke, Joseph A, "Holographic Visualization Of Acoustic Fields," J. Sound & Vib, 56(2), 167, 1978.
- 8. ---, "Visual Characteristics of Inhomogeneous Acoustic Waves," J. Sound & Vib, 70(2),267,1980.
- 9. ---, "Numerical Prediction of Supersonic Bulk-Cavitation Closure Pulses," Catholic University of America Technical Report to Office of Naval Research under Contract No. N00014-76-C-1020, January, 1981.
- 10. Cole, Robert H, UNDERWATER EXPLOSIONS, Princeton University Press, Princeton, N.J. (1948).
- 11. Costanzo, Frederick A, and John D. Gordon, "An Analysis of Bulk Cavitation in Deep Water," Underwater Explosive Research Division of the Naval Ship Research and Development Center, Portsmouth, Va. May, 1980.
- 12. Cushing, Vincent J, and William Losaw, "Hull Plate Deformation from Underwater Shock Waves," Engineering-Physics Co, Final Report to Underwater Explosive Research Division of David Taylor Model Basin under Contract No. N189(181)-56855A(X) (July, 1965).
- 13. ---, "Study of Bulk Cavitation and Consequent Water Hammer," Engineering-Physics Co. Final Report to ONR under contract Nonr-3389(00), 1961.
- 14. ---, George Bowden & Dean Reily, "Three-Dimensional Analysis of Bulk Cavitation," Engineering-Physics Co. Interim Report to ONR under contract NONR-3709(00), September 24, 1962.

- 15. ---, "On The Theory Of Bulk Cavitation," Engineering-Physics Co. Final Report to ONR under Contract No. Nonr-3709(00), December, 1969.
- 16. ibid, Ch. V, p43 et seq.
- 17. DeRuntz, J. A, T. L. Geers and C. A. Felippa, "The Underwater Shock Analysis Code, A Reference Manual," Lockheed Palo Alto Research Laberatory, Final Report to Defense Nuclear Agency under contract DNA-001-/ú-c-0285, February, 1978.
- 18. DiMaggio, F. L, I. S. Sandler and D. Rubin, "Uncoupling Approximation in Fluid-Structure Interaction Problems with Cavitation," Weidlinger Associates Interim Report to Defense Nuclear Agency under Contracts Nos. DNA 001-79-C-0078 and DNA 001-79-C-0256, February, 1980.
- 19. F. G. Friedlander, SOUND PULSES, Cambridge University Press, 1958.
- 20. Gaspin, J. B, and R. S. Price, "The Underpressure Field from Explosions in Water as Modified by Cavitation," NOLTR 72-103, 9 May 1972.
- 21. Geers, T. L, "Residual Potential and Approximate Methods for Three-Dimensional Fluid-Structure Interaction Problems," JASA, 49(5) (Part 2), 1505 (1971).
- 22. Haywood, J. H, "Response of and Elastic Cylindrical Shell to a Pressure Pulse," J. Mech. Appl. Math. XI (Part 2), 129 (1958).
- 23. Huang, H, G. C. Everstine and Y. F. Wang, "Retarded Potential Techniques for the Analysis of Submerged Structures Impinged by Weak Shock Waves," Computational Methods for Fluid-Structures Interaction Problems, ASME 26 (1977).
- 24. Hudson, G.E, and C.T. Taylor, "Time-Displacement Studies of Diaphragms Deformed by Explosive Loading," Underwater Explosion Research, Vol III, pp. 445-459, ONR, 1950.
- 25. Kennard, E. H, "Explosive Load on Underwater Structures as Modified by Bulk Cavitation," David Taylor Model Basin Report No. 511, 1943.
- 26. ---, "Cavitation in an Elastic Liquid," Phys. Rev. 63, 172, 1943.
- 27. ---, "The Effect of a Pressure Wave on a Plate or Diaphragm," Underwater Explosion Research, Vol III, pp. 9-106, ONR, 1950.
- 28. Kriebel, A. R, and J. S. Bechtel, "Hydrodynamic Data from Exploding Wires," URS Research Co Annual Report URS 679-6 to ONR under contract N0014-67-C-0451, April 1, 1970.

- 29. Mindlin, R. D, and H. H. Bleich, "Response of an Elastic Cylindrical Shell to a Transverse, Step Shock Wave," J. Appl. Mech. 20, 189, 1953.
- 30. Mnev, Y. N, and A. K. Pertsev, "Hydroelasticity of Shells," Translation Division, Foreign Technical Division, Wright-Patterson AFB, FTD-MT-24-119-71.
- 31. Murray, W. W, "Model Studies of Underwater Atomic Explosions on Ships, Part V, Results of Panel Tests," UERD Rept 3-58, 1948.
- 32. NAVSEA 0908-LP-000-0010, "Test Plan For Routine Shock Testing of Ships," Naval Ship Engineering Center, Hyattsville, Maryland, January, 1975.
- 33. Newton, R. E, "Effects of Cavitation on Underwater Shock Loading," Naval Postgraduate School Interim Report NPS-69-78-013 to Defense Nuclear Agency under MIPR 78-654, July, 1978.
- 34. ---, "Effects of Cavitation in Underwater Shock Loading," Naval Postgraduate School Final Report to Defense Nuclear Agency under Grant No. MIPR 79-608, April, 1980.
- 35. Officer, C. B, SOUND TRANSMISSION, McGraw-Hill, New york City, 1958.
- 36. Ranlet, D, F. L. DiMaggio, H. H. Bleich and M. L. Barron, "Elastic Response of Submerged Shells with Internally Attached Structures to Shock Loading," Computers and Structures, 7, 355, 1977.
- 37. Rehak, Margareta L, Frank L. DiMaggio and Ivan S. Sandler, "Interactive Approximations for a Cavitating Fluid around a Floating Structure," Computers and Structures, 21, NO.6, 1159, 1985.
- 38. ---, M, R. Smilowitz and R. Kagel, "The FRAM Cavitation Code," Weidlinger Associates Technical Report DNA-TR-86-179 to Defense Nuclear Agency under contract DNA-001-84-C-0001, 15 May 1986.
- 39. Reily, Dean, "Hydro Shock Experiment," Engineering-Physics Co. Final Report, ONR Contract No. N000 14-69-C-1238, 7 Jan. 1970.
- 40. Sachs, David A, "Underwater Shockwave Focusing at Caustics," Cambridge Acoustical Associates Final Report U-322-188 to ONR under contract N00014-66-C-0110, 31 Aug. 1969.
- 41. Schauer, Heinrich N, UERD Rept. 17-49, Nov, 1949.
- 42. ---, UERD Rept. 4-50, May, 1950.
- 43. Schultz, Michael E, and Vincent J. Cushing, "Surface Velocity Measurements at Mono Lake," Engineering-Physics Co. to ONR under contract No. N00014-66-C-0165 for Office of Naval Research, Washington, D.C, August, 1970.

- 44. Smilowitz, R, R. Kagel and I. Sandler, "Shock Response of a Partially Submerged Linearly Elastic Structure (User's Manual for the SRUE Code)," Weidlinger Associates Technical Report DNA-TR-86-25 to Defense Nuclear Agency under contract DNA-001-84-C-0001, 10 Jan. 1986.
- 45. Taylor, G. I, "The Pressure and Impulse of Submarine Explosion Waves on Plates," Underwater Explosion Research, Vol. I, pp. 115-1173, ONR, 1950.
- 46. Waldo, G. V, "A Bulk Cavitation Theory with a Simple Exact Solution," NSRDC, Washington, DC, Rpt 3010, April, 1969.
- 47. Walker, R. R, and J. D. Gordon, "A study of the Bulk Cavitation Caused by Underwater Explosions," DTMB Report 1896, 1966.
- 48. Wentzell, R. A, "Cavitation Due to Shock Pulses Reflected from the Sea Surface," JASA, 46, No.3 (part 2), 1969.

APPENDIX

BULK CAVITATION Z100 Basic Program

10 ' CAVIT.7CN BASIC VERSION 12 GOTO 40 20 CLS: LOCATE 4.1 22 PRINT = UERD/DINSRDC (DNA). 9 OCT, 1980 REVISION TO INCLUDE EFFECTS OF BOTTOM LAYER ACCELERATION BY J. CLARK (CUA). 23 DEC 1987 REVISION TO INCLUDE UPPER BOUNDARY DYNAMICS, AFTERFLOW AND CLOSURE MACH NUMBER BY V. CUSHING (CAT) 24 PRINT - 11 26 PRINT " 28 PRINT " PRINT " 30 PRINT " 32 PRINT " PRINT " 34 35 PRINT " 36 PRINT " 37 PRINT " 38 PRINT " 39 RETURN 5: N2 = 4: SFORM2 = STRING\$(N1,"#") + "." + STRING\$(N2,"#"): NN = N1+N2+1: SFORM1 = "\" + SPACE\$(NN-2) + "\" 122 N1 =500 ' INITIALIZATION 510 PATMOS = 14.7#144: RHO = 2: GEE = 32.2: CWAT = 5000 520 GAMMA = RHO#GEE: AINT = 2#RHO/PATMOS ' <---> PSI IN INTEGRAL (TOTAL IN LINE) 530 RSTART = 200 540 AIMP = RHOTCWAT 550 $144^{2}23976$: AL = 1.13: AM = $.0654^{4}.001$: AN = -.18HBX-1 AK

 >50
 AK
 =
 144'

 PARAMETERS
 554
 GOSUB
 20

 555
 INPUT
 "

 556
 INPUT
 "

 570
 W3
 =
 W1
 333

 574
 INPUT
 "
 580
 PRINT
 "

 580
 PRINT
 "
 500
 ND7
 =
 500

 OUTPUT FILE NAME : ", SOUT POUNDS OF HE : ", W BURST DEPTH : " PROGRAM IS NOW COMPUTING MAXIMUM CAVITY RADIUS" 590 NDZ = 500 1000 ' ENTRY POINT 1010 GOSUB 8000 1020 PRINT " 1022 INPUT " THE MAXIMUM CAVITY RADIUS IS : " R(0) 1020 PRINT " # RADIAL DATA POINTS AS N#; OR DR AS D#: ", S 1024 IF LEFT\$(S,1) = "N" THEN DELRAD = R(0)/VAL(MID\$(S,2)) 1026 IF LEFT\$(S,1) = "D" THEN DELRAD = VAL(MID\$(S,2)) 1030 PRINT " PROGRAM IS NOW COMPUTING VERTICAL BOUNDS": LIN = **ČSRLIN: GOSUB 9000** PROGRAM IS NOW COMPUTING CLOSURE": LIN = 1040 PRINT " LIN+1: GOSUB 10000 1044 PRINT " GOSUB 15000 1050 ' PLOT CLOSURE WAVE 1054 PRINT SPACE\$(N1-2); 1060 PRINT USING SFORM1; " R "; "Z-UPPER"; "Z-LOWER"; "Z-CLOSE"; "T-CLOSE"; "ZSDOT"; "ZBDOT"; "1/MACH" 1070 FOR I = 1 TO IMAX 1080 PRINT USING SFORM2; R(I); Z(I,1); Z(I,2); Z(I,3); T(I,3); ZS DOT(I); ZBDOT(I); RMACH(I) 1092 NEVT T 1044 PRINT " PROGRAM IS NOW WRITING TO FILE": 1100 SYSTEM 1200 END ----- SUBROUTINE LOWER(R,Z) 2000 '----2010 Z = 0: GOSUB 7000: FMEM = F: XMEM = Z 2020 Z = 2.5: IFLAG = 0: ICOUNT = 0

```
2030 GOSUB 7000: X = Z
2040 GOSUB 4000: Z = X
2050 IF IFLAG = 0 GOTO 2030
2070 RETURN
2080 '
2000 '

3000 '------ SUBROUTINE UPPER(R,Z)

3010 Z = 0: GOSUB 5000: FMEM = F: XMEM = Z ' 5000 REQUIRES R,Z

3020 Z = .1: IFLAG = 0: ICOUNT = 0

3030 GOSUB 5000: X = Z

3040 GOSUB 4000 : Z = X

3040 GOSUB 4000 : Z = X

3050 IF IFLAG = 0 GOTO 3030

3060 GOSUB 6500: T = TAR

3070 RETURN
3070 RETURN
3080 '
4000
                    ----- SUBROUTINE FOR F ---> O
4010 XNEW = X - F^*(X-XMEM)/(F-FMEM)
4020 XDIF = ABS(XNEW-X)
4030 XMIN = .05
4030 XMIN = .05
4040 IF (XDIF < XMIN) THEN IFLAG = 1
4050 \text{ XMEM} = X
4060 X = XNEW
4070 FMEM = F
4090 RETURN
4100
5000 '-----
5010 GOSUB 6000
                     ----- SUBROUTINE UPP1(R,Z,F)
5020 F = -PREL - AK*((W3/RA)_AL)
5030 RETURN
5040 '
                  ----- SUBROUTINE PRESS(R,Z,PREL)
6000
6050 PHYD = GAMMA *Z
6060 PREL = -(PINC + PATMOS + PHYD): UA = PREL/AIMP
6090 UZ = -UI*(D-Z)/RI + UA*(D+Z)/RA: UR = UI*R/RI + UA*R/RA
6100 RETURN
7070 RETURN
7080 '
8000 '-----
                    ----- SUBROUTINE TANPT(R3)
8010 R = RSTART
8030 GOSUB 3000: GOSUB 7000: RMEM = R: GMEM = F
8040 R = 2*R
8060 GOSUB 3000: GOSUB 7000: IFLAG = 0
8070 X = R: XMEM = RMEM: FMEM = GMEM: GOSUB 4000: R = X: RMEM = XMEM:
                GMEM = FMEM
8080 IF IFLAG = 0 GOTO 8060
8100 R(0) = R
8110 RETURN
8120
9000 '----- SUBROUTINE BOUNDS(R(0))
9010 R = 0: I = 1: FINAL = R(0) + .5*DELRAD
9020 LOCATE 16.1: PRINT "RADIAL COMPUTATION #: ": GOSUB 20000
PRINT COUNTER
9030 GOSUB 3000
9040 Z(I,1) = Z: T(I,1) = TAR
9050 GOSUB 2000
```

i.

1

9060 Z(I,2) = Z: T(I,2) = TAR: R(I) = R 9070 IF Z > Z2MAX THEN Z2MAX = Z 9090 R = R + DELRAD: I = I + 1 9100 IF R < FINAL GOTO 9020 9110 IMAX = I-1 9120 RETURN 9130 ' 10000 '------- SUBROUTINE CCURVE 10010 FOR I = 1 TO IMAX 10020 DZ = Z2MAX/NDZ 10030 R = R(I) 10040 GOSUB 20000 10060 ----- SET UP BOTTOM FACTORS 10100 10100 Z = Z(I,2)10120 GOSUB 6000 10130 TA = TAR: TE = TI: TI1 = TI0 10140 THETA = THETAI 10150 COEF1 = -PIO/AIMP: CONST1 = THETA/TI1: COEF2 = 1 - CONST1 10160 COEF3 = COEF1#THETA: COEF4 = CONST1 - 1: CONST2 = EXP(-TE/THETA) 10170 FACTVI = (D-Z)/RI: FACTVA = (D+Z)/RA 10180 FACTRI = R/RI: FACTRA = R/RA 10190 UAO = UA10210 Z = 0: GOSUB 6000: UZMEM = UZ 10220 TOTAL = 0: ZMEM = 0: ZB = 0: ZS = 0: ZSMEM = 0: TTMEM = TAR: IFLAG2 = 0 10230 GOSUB 10500: FMEM = F 10240 Z = DZ: ICOUNT = 0 10250 FOR CONVERGENCE 10340 Z = Z + DZ 10350 GOTO 10300 10360 GOSUB 10500 10370 X = Z: GOSUB 4000: Z = X 10380 IF IFLAG = 0 THEN ZMEM = XMEM: GOTO 10300 10390 GOSUB 11000 10400 IF I = 1 THEN RMACH(I) = 0: GOTO 10420 10410 DS = SQR((R-RLAST)_2 + (Z-ZLAST)_2): DT = TT - TLAST: RMACH(I) = DT + CWAT/DS DT CWAT/DS 10420 RLAST = R: TLAST = TT: ZLAST = Z 10430 Z(I,3) = Z: T(I,3) = TT: ZSDOT(I) = ZSDOT: ZBDOT(I) = ZBDOT: DELDOT(I) = ZSDOT - ZBDOT 10480 NEXT I 10490 RETURN 10500 ' ----- SET UP F FOR CLOSURE 10510 F = ZS - ZB10520 RETURN 10530 1 ----- START LOOP WITH SURFACE MOTION 11010 GOSUB 6000 11010 GOOD GOOD 11020 IF IFLAG2 = 1 THEN DZ = Z - ZMEM 11030 TOTAL = TOTAL + .5*(UZ + UZMEM)*DZ: UZMEM = UZ 11040 TF = AINT*(Z*UZ - TOTAL) ' TIME OF FLIGHT TF (SEC) AS FUNCTION OF SPALL THICKNESS Z (FEET) 11050 ZR = UR*TF 11060 ZS = TF*UZ + .5*GEE*TF*TF ' SURFACE DISPLACEMENT --- EQS. 14 AND BEGIN BOTTOM MOTION CALCULATION 11200 -11210 T = TT-TI1 11220 VI = $COEF1^{*}(COEF2^{*}EXP(-T/THETA) + CONST1)$ 11230 ZI = $COEF3^{*}((EXP(-T/THETA) - CONST2)^{*}COEF4 + (T-TE)/TI1)$

.

DISTRIBUTION LIST

DNA-TR-89-289

DEPARTMENT OF DEFENSE ASSISTANT TO THE SECRETARY OF DEFENSE ATTN: EXECUTIVE ASSISTANT DEFENSE INTELLIGENCE AGENCY ATTN: DB-4C3 ATTN: DB-6E1 ATTN: DB-6E2 C WIEHLE ATTN: RTS-2B DEFENSE NUCLEAR AGENCY ATTN: OPNS ATTN: SPSD ATTN: SPSD LCDR C NOFZIGER ATTN: SPWE ATTN: SPWE A FREDERICKSON ATTN: SPWE E TREMBA ATTN: SPWE LCDR M O'BRYANT 4 CYS ATTN: TITL DEFENSE NUCLEAR AGENCY ATTN: TDNV DEFENSE NUCLEAR AGENCY ATTN: TDNM ATTN: TDTT DEFENSE TECHNICAL INFORMATION CENTER 2 CYS ATTN: DTIC/FDAB FIELD COMMAND DEFENSE NUCLEAR AGENCY ATTN: FCPR THE JOINT STAFF ATTN: JKC (ATTN: DNA REP) ATTN: JKCS DEPARTMENT OF THE ARMY **DEP CH OF STAFF FOR OPS & PLANS** ATTN: DAMO-SWN ENGINEER STUDIES CENTER ATTN: SECURITY MANAGER HARRY DIAMOND LABORATORIES ATTN: SLCIS-IM-TL **U S ARMY CORPS OF ENGINEERS** ATTN: CERD-L

U S ARMY ENGR WATERWAYS EXPER STATION ATTN: C WELCH CEWES-SE-R ATTN: CEWES J K INGRAM ATTN: CEWES-CW D OUTLAW ATTN: CEWES-SD J G JACKSON, JR ATTN: J ZELASKO CEWES-SD-R ATTN: R WHALIN CEWES-ZT ATTN: RESEARCH LIBRARY ATTN: S HUGHES

U S ARMY FOREIGN SCIENCE & TECH CTR ATTN: AIFRTA **U S ARMY MATERIAL TECHNOLOGY LABORATORY** ATTN: DRXMR J MESCALL ATTN: TECHNICAL LIBRARY U S ARMY MISSILE COMMAND/AMSMI-RD-CS-R ATTN: AMSMI-RD-CS-R **U S ARMY NUCLEAR & CHEMICAL AGENCY** ATTN: MONA-NU D BASH **U S ARMY STRATEGIC DEFENSE COMMAND** ATTN: CSSD-SA-EV ATTN: CSSD-SL **U S ARMY WAR COLLEGE** ATTN: LIBRARY USA SURVIVABILITY MANAGMENT OFFICE ATTN: SLCSM-SE J BRAND **DEPARTMENT OF THE NAVY** DAVID TAYLOR RESEARCH CENTER ATTN: CODE 11 ATTN: CODE 172 ATTN: CODE 173 ATTN: CODE 1740.1 ATTN: CODE 1740.4 ATTN: CODE 1740.5 ATTN: CODE 1740.6 ATTN: CODE 1770.1 ATTN: CODE 2740 MAGTE WARFIGHTING CENTER ATTN: DO91 J HARTNEADY MARINE CORPS ATTN: CODE POR-21 NAVAL COASTAL SYSTEMS CENTER ATTN: CODE 7410 NAVAL DAMAGE CONTROL TRAINING CENTER ATTN: COMMANDING OFFICER NAVAL ELECTRONICS ENGRG ACTVY, PACIFIC ATTN: CODE 250 NAVAL EXPLOSIVE ORD DISPOSAL TECHNOLOGY CENTER ATTN: CODE 90 J PETROUSKY NAVAL POSTGRADUATE SCHOOL ATTN: B K WOEHLER ATTN: CODE 1424 LIBRARY NAVAL RESEARCH LABORATORY ATTN: CODE 2627 ATTN: CODE 5100 ATTN: CODE 6380

NAVAL SEA SYSTEMS COMMAND ATTN: PMS 421B A COTE, JR ATTN: SEA-033 ATTN: SEA-08 ATTN: SEA-09G53 ATTN: SEA-323 ATTN: SEA-55X ATTN: SEA-55X1 ATTN: SEA-55Y ATTN: SEA-9931G NAVAL SURFACE WARFARE CENTER ATTN: CODE H21 ATTN: CODE R14 ATTN: CODE R15 ATTN: CODE U401 M KLEINERMAN NAVAL SURFACE WARFARE CENTER ATTN: CODE K42 R ROBINSON ATTN: TECHNICAL LIBRARY NAVAL WEAPONS CENTER ATTN: CODE 3263 J BOWEN NAVAL WEAPONS EVALUATION FACILITY ATTN: CLASSIFIED LIBRARY **NEW LONDON LABORATORY** ATTN: CODE 4492 J KALINOWSKI ATTN: CODE 4494 J PATEL ATTN: TECH LIBRARY OFFICE OF CHIEF OF NAVAL OPERATIONS ATTN: NOP 098T8 ATTN: NOP 223 ATTN: NOP 225 ATTN: NOP 37 ATTN: NOP 605D5 ATTN: NOP 957E ATTN: OP 03EG ATTN: OP 21 ATTN: OP 654 ATTN: OP 73 ATTN: OP 02 OFFICE OF NAVAL RESEARCH ATTN: CODE 1132SM THEATER NUCLEAR WARFARE PROGRAM OFC ATTN: PMS-42332C US NAVAL ACADEMY ATTN: LIBRARY **DEPARTMENT OF THE AIR FORCE** AIR FORCE INSTITUTE OF TECHNOLOGY/EN ATTN: COMMANDER **HEADOUARTERS USAF/IN** ATTN: IN HO USAF/CCN ATTN: AFCCN

ROME AIR DEVELOPMENT CENTER, AFSC ATTN: COMMANDER

USAF/LEEEU

ATTN: LEEE

WEAPONS LABORATORY ATTN: NTE ATTN: NTED G GOODFELLOW ATTN: NTED R HENNY ATTN: NTES ATTN: WL/SUL

•

,

DEPARTMENT OF ENERGY

DEPARTMENT OF ENERGY OFFICE OF MILITARY APPLICATIONS ATTN: OMA/DP-225

LAWRENCE LIVERMORE NATIONAL LAB ATTN: C E ROSENKILDE ATTN: D MAGNOLI

LOS ALAMOS NATIONAL LABORATORY ATTN: T DOWLER ATTN: REPORT LIBRARY

MARTIN MARIETTA ENERGY SYSTEMS INC ATTN: CIVIL DEF RES PROJ ATTN: CENTRL RESRCH LIB 4500N

SANDIA NATIONAL LABORATORIES ATTN: DIV 9311 J S PHILLIPS ATTN: TECH LIB 3141

OTHER GOVERNMENT

CENTRAL INTELLIGENCE AGENCY ATTN: OSWR/NED

DEPARTMENT OF THE INTERIOR ATTN: D RODDY

U S NUCLEAR REGULATORY COMMISSION ATTN: R WHIPP FOR L SHAO

DEPARTMENT OF DEFENSE CONTRACTORS

AMI RESEARCH ATTN: V GODINO

APPLIED RESEARCH ASSOCIATES, INC ATTN: R FRANK

BDM INTERNATIONAL INC ATTN: E DORCHAK

BOEING TECHNICAL & MANAGEMENT SVCS, INC ATTN: R SCHMIDT

CALIFORNIA INSTITUTE OF TECHNOLOGY ATTN: T AHRENS CALIFORNIA RESEARCH & TECHNOLOGY, INC ATTN: K KREYENHAGEN ATTN: LIBRARY ATTN: M ROSENBLATT ATTN: S SCHUSTER

CALIFORNIA RESEARCH & TECHNOLOGY, INC ATTN: J THOMSEN ATTN: R ENGLAND

CALIFORNIA, UNIVERSITY AT SAN DIEGO (A-030) ATTN: W VAN DORN

COLUMBIA UNIVERSITY ATTN: F DIMAGGIO

CUSHING ASSOCIATES, INC 2 CYS ATTN: V J CUSHING

DENVER COLORADO SEMINARY UNIVERSITY OF ATTN: J WISOTSKI

ENGINEERING METHODS & APPLICATIONS INC ATTN: DAVID DIVOKY

KAMAN SCIENCES CORP ATTN: L MENTE ATTN: LIBRARY ATTN: R RUETENIK

KAMAN SCIENCES CORP ATTN: F SHELTON ATTN: LIBRARY B KINSLOW

KAMAN SCIENCES CORP ATTN: DASIAC ATTN: E CONRAD

KAMAN SCIENCES CORPORATION ATTN: DASIAC

KARAGOZIAN AND CASE ATTN: J KARAGOZIAN

LOCKHEED MISSILES & SPACE CO, INC ATTN: PHILIP UNDERWOOD

LOCKHEED MISSILES & SPACE CO, INC ATTN: TECH INFO CTR

MCDONNELL DOUGLAS CORPORATION ATTN: R HALPRIN

MIAMI, UNIVERSITY OF ATTN: B LEMEHAUTE ATTN: S WANG

NEW MEXICO ENGINEERING RESEARCH INSTITUTE ATTN: J JARPE ATTN: N BAUM

PACIFIC-SIERRA RESEARCH CORP ATTN: H BRODE PHYSICS APPLICATIONS, INC ATTN: DOCUMENT CONTROL **R & D ASSOCIATES** ATTN: B LEE **R & D ASSOCIATES** ATTN: E FURBEE ATTN: J WEBSTER RAND CORP ATTN: B BENNETT ROSENSTIEL SCHOOL OF MARINE ATTN: APPL MAR PHYSICS-GROSVENOR ATTN: APPL MAR PHYSICS-MSC124 S-CUBED ATTN: K D PYATT, JR ATTN: R SEDGEWICK SCIENCE APPLICATIONS INTL CORP ATTN: DR M MCKAY ATTN: H WILSON ATTN: TECHNICAL REPORT SYSTEM SCIENCE APPLICATIONS INTL CORP ATTN: G BINNINGER SCIENCE APPLICATIONS INTL CORP ATTN: R ALLEN SCIENCE APPLICATIONS INTL CORP ATTN: J R BRITT SRI INTERNATIONAL ATTN: D KEOUGH ATTN: J COLTON ATTN: M SANAL ATTN: P DE CARLI **TELEDYNE BROWN ENGINEERING** ATTN: J RAVENSCRAFT TRW INC ATTN: TECH INFO CTR **TRW SPACE & DEFENSE SECTOR SPACE** ATTN: OUT6 W WAMPLER WEIDLINGER ASSOC, INC ATTN: J ISENBERG WEIDLINGER ASSOCIATES, INC ATTN: T DEEVY WEIDLINGER ASSOCIATES, INC ATTN: M BARON WESTINGHOUSE ELECTRIC CORP ATTN: D BOLTON