

2

TECHNICAL REPORT BRL-TR-3192

# BRL

AD-A237 621

## VARIATIONAL METHOD IN THE STATISTICAL THEORY OF TURBULENCE

G. DOMOKOS  
S. KOVESI-DOMOKOS  
C. K. ZOLTANI

DTIC  
ELECTE  
FEB 11 1991  
S B D

JANUARY 1991

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED.

U.S. ARMY LABORATORY COMMAND

BALLISTIC RESEARCH LABORATORY  
ABERDEEN PROVING GROUND, MARYLAND

91 2 08 023

## NOTICES

Destroy this report when it is no longer needed. DO NOT return it to the originator.

Additional copies of this report may be obtained from the National Technical Information Service, U.S. Department of Commerce, 5285 Port Royal Road, Springfield, VA 22161.

The findings of this report are not to be construed as an official Department of the Army position, unless so designated by other authorized documents.

The use of trade names or manufacturers' names in this report does not constitute indorsement of any commercial product.

UNCLASSIFIED

REPORT DOCUMENTATION PAGE			Form Approved OMB No 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.				
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE January 1991	3. REPORT TYPE AND DATES COVERED Final Jan 88 - Jun 90		
4. TITLE AND SUBTITLE Variational Method in the Statistical Theory of Turbulence			5. FUNDING NUMBERS STAS # DAAL03-86-D-0001 1L161102AH43	
6. AUTHOR(S) G. Domokos, S. Kovesi-Domokos, and C.K. Zoltani*				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)			8. PERFORMING ORGANIZATION REPORT NUMBER	
9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES) USA Ballistic Research Laboratory ATTN: SLCBR-DD-T Aberdeen Proving Ground, MD 21005-5066			10. SPONSORING / MONITORING AGENCY REPORT NUMBER BRL-TR-3192	
11. SUPPLEMENTARY NOTES *Names appear alphabetically.				
12a. DISTRIBUTION / AVAILABILITY STATEMENT Approved for Public Release - Distribution Unlimited			12b. DISTRIBUTION CODE	
13. ABSTRACT (Maximum 200 words) A new approach to the calculation of turbulent flows is presented. Variational principles are introduced into the statistical theory of turbulence enabling the determination of correlation functions by extremizing certain functionals. To demonstrate the efficiency of the technique a sample calculation of a mean flow profile and certain two point correlation functions of a cylindrically symmetric free jet were carried out using the Rayleigh-Ritz method. A judicious choice of the trial functions leads to the determination of the correlation functions with a modest amount of computation. The results are in reasonable agreement with experimental data.				
14. SUBJECT TERMS Two Phase Flow, Turbulence, Statistical Turbulence			15. NUMBER OF PAGES 33	
			16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT UNCLASSIFIED	18. SECURITY CLASSIFICATION OF THIS PAGE UNCLASSIFIED	19. SECURITY CLASSIFICATION OF ABSTRACT UNCLASSIFIED	20. LIMITATION OF ABSTRACT SAR	

INTENTIONALLY LEFT BLANK.

# TABLE OF CONTENTS

	<u>Page</u>
TABLE OF CONTENTS.....	iii
1. INTRODUCTION.....	1
2. REVIEW OF THE STATISTICAL APPROACH.....	2
3. THE MATHEMATICAL APPROACH.....	4
3.1 Incompressible Flow: Potential and Symmetries.....	4
3.2 Variational Principles for the Correlation Functions.....	7
4. CALCULATION OF A CYLINDRICALLY SYMMETRIC JET.....	10
5. DISCUSSION.....	17
REFERENCES.....	23
DISTRIBUTION LIST.....	25

DTIC  
COPY  
INSPECTED

<b>Accession For</b>	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By	
Distribution/	
Availability Codes	
Dist	Avail and/or Special
A-1	

INTENTIONALLY LEFT BLANK.

## 1. INTRODUCTION

The current generation of interior ballistic codes rely on turbulence models which incorporate little if any of the underlying physics of the gas-particle motion in a gun tube. Indeed, more often than not, the fictitious length scales and adjustable constants employed are based on experiments which have no relationship to the flow being modeled. The result at best is post predictive in nature and no claim to physical representation can be made. To put turbulence sub-models in a ballistic setting on a more realistic footing, this study was undertaken. A new mathematical approach to the problem is presented and results for the single phase, turbulent case are given. In a companion paper the flow in a two-phase jet is analyzed and compared to data obtained at BRL.

Turbulence is prevalent in practically all naturally occurring flow processes and has challenged, without adequate resolution, the best scientific minds for over a century. The problem is especially hard due to the very large number of coupled degrees of freedom involved. Hence the description of such fluid flows cannot proceed *via* usual techniques of approximating the system by a linear one and using perturbation techniques around the linearized limit. Despite the existence of an enormous body of theoretical literature on the subject, there is still no "standard" theoretical treatment of fully developed turbulence. In our previous works we took a statistical approach to this problem (Domokos, Domokos-Kovesi and Zoltani 1988), following the pioneering works of Martin, Siggia, Rose (1973) and De Dominicis and Peliti (1978). In this approach one considers the classical equations of motion of a fluid, *viz.* the Navier-Stokes equations, perturbed by a random force. From the physical point of view, the random force represents the fluctuations in the fluid; its presence is necessary in order to avoid unstable, non-turbulent solutions of the equations of motion. The statistics of the random force, in turn, generates a statistical distribution of the components of the flow velocity. The correlation functions computed from the latter distribution can be compared directly with the results of measurements. An application of the technique developed by these authors to the case of turbulent channel flow led to a reasonable agreement with the available experimental data, cf. Burgett (1989). The approach adopted there consisted of a combination of analytical and numerical methods: an analytic approach was followed and numerical computations were used only for the computation of Fourier transforms, integrations, etc. While this approach led to a considerable reduction of computational complexity, it became clear that a straightforward application of those techniques to problems like fluid flow in jets would lead to an unreasonably large amount of computing time.

There are basically two features of the approach which make the computations difficult. First, there is no easy way available to satisfy the constraint imposed by the equation of continuity on the various velocity correlation functions; second, the approximation methods used are essentially perturbative in nature, hence, one has to labor very hard in order to achieve a reasonable accuracy.

In this work we describe some further developments in the statistical theory of fully developed turbulence which are designed to avoid the difficulties outlined above. First, we describe a technique by means of which the equation of continuity can be automatically satisfied. (We explicitly describe the procedure for the case when the fluid may be regarded an incompressible one, although the method can be generalized to arbitrary, compressible flows.) Second, we develop a variational approach to the computation of correlation functions. This enables us to avoid perturbative approaches altogether.

The paper is organized as follows. In the next section we briefly review the statistical approach to turbulent flows as described in Domokos, Domokos–Kovesi and Zoltani (1988). In Sec. 3 we introduce vector potentials which enable us to satisfy the equation of continuity identically. We also discuss the method of imposing various symmetry requirements on the correlation functions. We work out explicitly the constraints imposed by the requirement of cylindrical symmetry. The variational approach is described in Sec. 3.2. A sample calculation is presented in Sec. 4: some correlation functions of an axisymmetric jet are calculated by means of the technique developed here. Sec. 5 contains a discussion of the results.

## 2. REVIEW OF THE STATISTICAL APPROACH

Throughout this paper, a condensed notation is used. We consider a class of systems described by a classical equation of motion of the general form:

$$\partial_t X + F[X] = f(x). \quad (2.1)$$

Here  $X$  stands for an element of the vector space of dynamical variables.  $F[X]$  is an autonomous map of the vector space upon itself;  $x$  stands for both spatial coordinates and the time variable,  $x = (\mathbf{x}, t)$ . Finally,  $f(x)$  represents a Gaussian random force acting upon

the system; it plays the role of a disordering field, driving the system described by eq. (2.1) away from a non-chaotic behavior. Specifically, if the system is an incompressible fluid described by the Navier-Stokes equations,  $X$  is identified with the velocity field,  $\mathbf{u}$ . The map  $F[X]$  reads:

$$F[\mathbf{u}]^i = (\mathbf{u}^s \partial_s) \mathbf{u}^i + \partial^i p - \nu \nabla^2 \mathbf{u}^i, \quad (2.2)$$

with the velocity field satisfying the constraint,  $\nabla \cdot \mathbf{u} = 0$ . In that case, the quantity  $p$  (the pressure divided by the density), is not an independent dynamical variable: it can be explicitly expressed in terms of  $\mathbf{u}$ . Due to the constraint upon  $\mathbf{u}$ , we are free to assume that the perturbing random force is solenoidal.

The quantity playing the central role in theories of this type is the generating functional of the correlation functions. It can be expressed in terms of the probability distribution of the random force as follows. Let  $K$  stand for the correlation operator of the random force. Then the generating functional of the correlation functions of  $X$  is given by the functional integral:

$$Z[j] = \int DX \exp - [((\partial_t X - F), K(\partial_t X - F)) + (j, X)], \quad (2.3)$$

where  $(\cdot, \cdot)$  stands for a scalar product over the vector space, involving integration over space-time variables and summation over tensor indices, whereas  $j$  is an arbitrary function: the functional derivatives of  $Z$  with respect to  $j$  give the correlation functions. The cumulants are generated by the functional  $W = -\ln Z$ . Letting  $f$  to be a white noise, *viz.*  $K(x_1, x_2) = k \delta(t_1 - t_2) \delta^3(\mathbf{x}_1 - \mathbf{x}_2)$  (where  $k$  is a constant) is often a satisfactory choice.

The functional weight of integration,  $DX$ , is proportional to the (infinite) determinant,

$$\text{Det} (\partial_t - \delta F / \delta X). \quad (2.4)$$

It was shown by Domokos, Domokos-Kovesy and Zoltani (1988) that the latter can be expressed in terms of a functional integral over Fadeev-Popov ghosts (Itzykson and Zuber

1980). The central questions are: what is a satisfactory implementation of the constraints imposed upon  $X$  (in the case of an incompressible fluid flow, the constraint  $\nabla \cdot \mathbf{u} = 0$ ) as well as the development of suitable approximation techniques for the computation of  $Z$  (or its functional derivatives).

### 3. THE MATHEMATICAL APPROACH

3.1 Incompressible Flow: Vector Potentials and Symmetries. An incompressible flow is characterized by the fact that the velocity field is solenoidal. Any solenoidal vector field is obtained as the curl of a vector potential. Hence, we write,  $\mathbf{u} = \nabla \times \mathbf{A}$ , where  $\mathbf{A}$  is the vector potential. It is also well known that, given the velocity field, the vector potential is determined only up to the gradient of an arbitrary scalar function:  $\mathbf{A}$  and  $\mathbf{A} + \nabla H$  give rise to the the same velocity field. (This is called the gauge freedom in electromagnetic theory: we adopt the same terminology here.) The scalar function,  $H$  can be chosen so as to simplify the problem of determining the vector potential. The equation satisfied by the vector potential is obtained by substituting the relationship,  $\mathbf{u} = \nabla \times \mathbf{A}$  into the Navier–Stokes equation. This is straight forward and we do not reproduce the result here. Correspondingly, in the statistical theory discussed by Domokos, Domokos–Kovesi and Zoltani (1988) and briefly reviewed in the preceding section, one first determines the correlation functions of the vector potential; the velocity correlation functions are then obtained by taking the curl with respect to every argument.

It is worth remarking that this way of satisfying the incompressibility constraint is not the only possibility: it would be possible to take the constraint into account by means of the Fadeev–Popov method, see Itzykson and Zuber (1980). However, the method described here is simpler and leads to results more quickly than any other approach we know of. This is due to the gauge freedom just described.

In what follows, we use a gauge such that the component of the vector potential along the mean flow vanishes. ("Axial gauge".) On denoting the mean flow by  $\mathbf{U}$ , the condition to be satisfied is:  $\mathbf{U} \cdot \mathbf{A} = 0$ . This can always be achieved. In fact, let  $\mathbf{B}$  be a vector potential which reproduces the velocity field, but its projection onto  $\mathbf{U}$  is not necessarily zero. Then, in order to satisfy  $\mathbf{U} \cdot \mathbf{A} = 0$ , with  $\mathbf{A} = \mathbf{B} + \nabla H$ , the scalar  $H$  has to satisfy the differential equation,

$$\mathbf{U} \cdot \mathbf{B} + \mathbf{U} \cdot \nabla H = 0. \quad (3.1.1)$$

There remains a residual gauge freedom, *viz.* a function  $h$  satisfying  $U \cdot \nabla h = 0$  can always be added to any solution of (3.1).

Any symmetry requirement should be now imposed upon the correlation functions of the vector potentials instead of the correlation functions of the velocity field itself. A symmetry of a flow means that the correlation functions are *invariant* under a subgroup of the Euclidean group. For instance, homogeneous turbulence means invariance under translations, hence, the  $n$ -point correlation function of the vector potential depends on the  $n-1$  coordinate differences only, not on the coordinates themselves. Likewise, in the case of isotropic turbulence, the correlation functions of order  $2n$  are proportional to the  $n$ -fold direct product of the metric tensor with itself, whereas correlation functions of odd order vanish.

Let us now concentrate on the two point correlation function; the construction of the general  $n$ -point correlation function proceeds in a similar fashion. For the sake of definiteness, we choose a coordinate system such that its third axis coincides with the direction of the mean flow,  $U$ . Due to the fact that the relationship between the vector potential and the velocity field is a linear one, a Reynolds decomposition of the vector potential leads to one of the velocity field. Let us write,

$$\begin{aligned} \mathbf{A} &= \langle \mathbf{A} \rangle + \mathbf{a}, \\ \mathbf{U} &= \nabla \times \langle \mathbf{A} \rangle, \\ \langle \mathbf{a} \rangle &= 0, \\ \mathbf{u}' &= \nabla \times \mathbf{a}, \end{aligned} \tag{3.1.2}$$

where  $\mathbf{u}'$  stands for the fluctuating part of the velocity field and  $\langle \dots \rangle$  denotes, as usual, the expectation value of a quantity. In an axial gauge we have:

$$A_3 = a_3 = 0, \tag{3.1.3}$$

so that only the components of the vector potential transverse to the mean flow are nonvanishing. (These components are denoted by subscripts/superscripts in capital letters,  $A, B$ , etc.)

The two-point correlation function of the vector potential  $\mathbf{a}(\mathbf{x}, t)$  can be decomposed in a basis of second rank tensors in the plane perpendicular to the mean flow. A basis of

such tensors consists of three elements. Correspondingly, the two-point correlation function can be written as follows:

$$Z_{AB} \equiv \langle a_A(\mathbf{x}_1, t_1) a_B(\mathbf{x}_2, t_2) \rangle = \delta_{AB} Z_1 + \epsilon_{AB} \epsilon_{RS}^{x_1 x_2} Z_2 + \frac{1}{2}(x_{1A} x_{2B} + x_{2A} x_{1B}) Z_3, \quad (3.1.4)$$

where  $\delta_{AB}$  and  $\epsilon_{AB}$  are the Kronecker and Levi-Civita tensors, respectively. The functions  $Z_1$ ,  $Z_2$  and  $Z_3$  are invariant under rotations around the third axis and they are symmetric functions of their arguments,  $(\mathbf{x}_1, t_1)$  and  $(\mathbf{x}_2, t_2)$ . (The antisymmetric product of the two coordinates has been inserted in front of  $Z_2$  so as to satisfy the permutation symmetry of the correlation function with all three invariant functions being symmetric.)

Thus we found that the two point correlation tensor of an incompressible fluid has three independent components only. Cylindrical symmetry of the flow further reduces the number of independent elements to two. In fact, cylindrical symmetry means that the correlation function is invariant under rotations around the third axis; the tensors  $\epsilon$  and  $\delta$  are the only invariant ones under such rotations; hence  $Z_3 = 0$ .

We now give the explicit expressions of the velocity correlation functions for a flow of cylindrical symmetry.

$$\begin{aligned} G_{AB} &= \delta_{AB} \partial_{13} \partial_{23} Z_1, \\ G_{A3} &= -\partial_{13} [x_{1A} Z_2 + \partial_{2A} Z_1], \\ G_{33} &= \delta^{RS} \partial_{1R} \partial_{2S} Z_1 + [2 + x_{1R} \partial_{1R} + x_{2R} \partial_{2R}] Z_2. \end{aligned} \quad (3.1.5)$$

The notation used in this and subsequent equations is the following. The first subscript denotes the argument in the correlation function; the second subscript or superscript refers to the vector component. The summation convention is used throughout.

Cylindrically symmetric turbulent flows of incompressible fluids have been, of course, discussed previously, cf. Batchelor (1946), Chandrasekhar (1950), Trevino (1982). Where

they overlap, our results agree with those of these works. The present approach is, however, more general, since it can be applied to flows of arbitrary symmetry and it is more explicit than the treatment of Trevino (1982).

Finally, we just mention that the present construction can be generalized in a straight forward fashion to compressible flows where both the velocity field and the density field are dynamical variables. In that case one has to introduce two vector potentials with a correspondingly enlarged group of gauge transformations; this problem will be discussed elsewhere.

3.2 Variational Principles for the Correlation Functions. In this Section we return to the condensed notation used in Sec. 2. Consider the expression of the generating functional of the cumulants, as quoted there. We have:

$$W[j] = - \ln \int DX \exp - [((\partial_t - F), K(\partial_t - F)) + (j, X)], \quad (3.2.1)$$

The averages of the various quantities are given by functional derivatives of  $W$ ,

$$G(1) \equiv \langle X(x_1) \rangle = \frac{\delta W}{\delta j(x_1)},$$

$$G(1,2) \equiv \langle X(x_1) X(x_2) \rangle = - \frac{\delta^2 W}{\delta j(x_1) \delta j(x_2)} + G(1)G(2), \quad (3.2.2)$$

etc. Instead of the functional argument  $j$ , we can now introduce  $G(1)$  as a functional argument, by means of a Legendre transformation in function space. Let us define a new functional,

$$S_1 = W - (j, G_1). \quad (3.2.3)$$

One readily verifies with the help of (3.2.1) that  $S_1$  is a functional of  $G(1)$ , its first functional derivative being given by the expression:

$$\frac{\delta S_1}{\delta G(1)} = -j(1) \quad (3.2.4)$$

The functional  $S_1$  can be determined from a functional differential equation obtained by combining eqs. (3.2.1) thru (3.2.4). This gives:

$$\begin{aligned} S_1 - (G_1, \frac{\delta S_1}{\delta j}) \\ = -\ln \int DX \exp - [((\partial_t X - F), K(\partial_t X - F)) - (\frac{\delta S_1}{\delta j}, X)]. \end{aligned} \quad (3.2.5)$$

The differential equation can be solved by iteration; one usually takes as a zeroth approximation the functional:

$$S_1^0 = S[G_1] \equiv ((\partial_t G_1 - F[G_1]), K(\partial_t G_1 - F[G_1])) \quad (3.2.6)$$

The principal advantage of the functional  $S_1$  is that it becomes stationary if the external "source",  $j$ , is put equal to zero, see eq. (3.2.4). hence, one may determine  $S_1$  up to a certain accuracy, but thereafter a calculation of  $G_1$  does not have to rely upon any perturbative approximation scheme. This procedure can be generalized if one is interested in determining the higher order correlation functions as well as the average of the dynamical variable itself. We outline the procedure for a scheme aimed at determining the functions  $G_1$  and  $G_2$ . It is necessary to introduce a bilinear source,  $h(1,2)$ , in addition to  $j$ , *viz.*

$$W[j,h] = -\ln \int DX \exp - [((\partial_t X - F), K(\partial_t X - F)) + (j, X) + (X, hX)] \quad (3.2.7)$$

Similarly to the procedure outlined above, one now performs a double Legendre transform in order to obtain a functional,  $S_2[G_1, G_2]$ :

$$S_2 = W[j,h] - (G_1, j) - \text{Tr}(hG_2), \quad (3.2.8)$$

where the trace is understood in the operator sense. One readily verifies the relations,

$$\begin{aligned}\frac{\delta S_2}{\delta G_2} &= -h, \\ \frac{\delta S_2}{\delta G_1} &= -j.\end{aligned}\tag{3.2.9}$$

Thus,  $S_2$  is stationary in both  $G_1$  and  $G_2$  in the limit of vanishing sources. The equation obeyed by the functional  $S_2$  can be determined in the same way as eq. (3.2.5). We merely quote the result:

$$\begin{aligned}S_2 - (G_1, \frac{\delta S_2}{\delta G_1}) - \text{Tr}(G_2 \frac{\delta S_2}{\delta G_2}) \\ = -\ln \int DX \exp - [((\partial_t X - F), K(\partial_t X - F)) - (\frac{\delta S_2}{\delta G_1}, X) - (X, \frac{\delta S_2}{\delta G_2} X)].\end{aligned}\tag{3.2.10}$$

Just as for the equation obeyed by  $S_1$ , the only known method of solving (3.2.10) is by iteration. We give here the first approximation to  $S_2$ :

$$S_2^1 = S[G_1] + \text{Tr}(\frac{\delta^2 S}{\delta G_1 \delta G_1} G_2) - \text{Tr}(\ln G_2).\tag{3.2.11}$$

(This form of  $S_2^1$  is obtained by changing to an integration variable,  $Y = X - G_1$  in eq. (3.2.10); thereafter  $S[Y + G_1]$  is expanded in powers of  $Y$  up to quadratic terms. The functional differential equation is then solved by quadrature.)

Functionals of the type (3.2.10) were first used by De Dominicis and Martin (1964) and Domokos and Suranyi (1964); the differential equations obeyed by such functionals was obtained by Cornwall, Jackiw and Tomboulis (1974). In the papers just quoted, it is also shown that higher order correlation functions can be computed by the application of either one of the variational principles described above. In this work, however, we concentrate upon the average flow (corresponding to  $G_1$ ) and the two point correlation function. For this purpose, the use of the variational principle based on the functional  $S_2$  is best suited.

#### 4. CALCULATION OF A CYLINDRICALLY SYMMETRIC JET

As an illustration of the technique developed above, we now perform a computation of some properties of a free jet, with a cylindrically symmetric mean flow. We use the variational principle (3.2.8) and compute the functional  $S_2$  in the first approximation, cf. eq. (3.2.11). The power of the variational principle lies in the fact that one can use a Rayleigh–Ritz method in order to minimize  $S_2$ . This leads to a considerable reduction in the complexity of computation: instead of solving coupled partial differential equations for the correlation functions, one tries to guess the form of the vector potential for the mean flow and of the functions  $Z_1$  and  $Z_2$  in (3.1.5), leaving a few free parameters. On substituting this form into the expression of  $S_2$ , the problem reduces to the computation of integrals and the minimization of the expression so obtained as a function of the parameters. The success of such an approach depends on finding a reasonable functional form of  $\langle A \rangle$  and of  $Z_1$  and  $Z_2$ . This can be accomplished only by a judicious use of one's physical intuition and by trial and error.

In this calculation we want to simplify the calculation as much as possible: we guess some simple functional forms which could reproduce the main features of the experimental data.

Let us examine now the Navier–Stokes operator entering the expression of the functional  $S_2$ :

$$N^i[\mathbf{u}] = \partial_t^i u^i - \nu \nabla^2 u^i + (\mathbf{u}^s \partial_s) u^i + \partial^i p \quad (4.1)$$

We are interested in fully developed turbulence: that means that it is described by a stationary ensemble. Hence the term containing the time derivative can be omitted from (4.1). Next we notice that if one introduces dimensionless quantities (as we shall do in what follows), the coefficient of the viscous term,  $\nabla^2 u$ , becomes proportional to  $1/\text{Re}$ , where  $\text{Re}$  stands for the Reynolds number. Hence, for flows of large Reynolds numbers ( $\text{Re} \approx 10^4$ , say), it is safe to omit the viscous term too, as long as we are interested in the behavior of the fluid on scales substantially larger than the dissipation scale. Thus, we are going to work with the truncated Navier–Stokes operator,

$$\mathbf{n}^i[\mathbf{u}] + \partial^i p, \quad (4.2)$$

with

$$\mathbf{n}^i[\mathbf{u}] = u^s \partial_s^i u^i.$$

The next task is to eliminate the pressure from the generating functional. To this end, we regard the pressure as one of the fluctuating dynamical variables and carry out the functional integration over it explicitly. Using an operator notation as before, we have to compute a functional integral of the form:

$$\int \mathcal{D}p \exp-\frac{i}{2}[\mathbf{n} + \nabla p] K [\mathbf{n} + \nabla p]. \quad (4.3)$$

This is a standard Gaussian integral. The result of the computation, to be inserted into the expression of the functional  $S_2$ , is the following:

$$\int \mathcal{D}p \exp-\frac{i}{2}[\mathbf{n} + \nabla p] K [\mathbf{n} + \nabla p] = \exp-\frac{i}{2}\mathbf{n} K^T \mathbf{n}, \quad (4.4)$$

with

$$K^T = K - (\nabla K)^{\sim} [\nabla K \nabla + \alpha^2]^{-1} \nabla K, \quad (4.5)$$

where  $\sim$  denotes the transpose of an operator.

In the last equation we inserted a constant  $\alpha^2$  in order to make the inverse of the operator  $\nabla K \nabla$  well defined on large length scales (equivalently, at small wave numbers). Evidently,  $K^T$  is the transverse part of  $K$ ,  $\nabla K^T = 0$ . We now have to make a physical assumption about the correlation operator,  $K$ . We argue that the stirring force should point in the direction of the mean flow: this is both intuitively plausible and it is the simplest way of satisfying the requirement of cylindrical symmetry of the problem. Otherwise, we assume the the correlation to be of short range. Hence we take, (cf. Sec. 2):

$$K_{ij}(\mathbf{x}, \mathbf{x}') = k \delta_{i3} \delta_{j3} \delta^3(\mathbf{x} - \mathbf{x}'). \quad (4.6)$$

In order to obtain a finite result even in the limit  $\alpha \rightarrow 0$ , we have to assume that  $k$  is proportional to  $1/\alpha$ . With this, the transverse part of  $K$  becomes:

$$K_{ij}^T(\mathbf{x}, \mathbf{x}') = a e^{-\alpha|z-z'|} \delta^2(\mathbf{x}^A - \mathbf{x}'^A) \delta_{i3} \delta_{j3}, \quad (4.7)$$

where  $a$  is a constant independent of  $\alpha$ . At the end of the calculation one may take  $\alpha \rightarrow 0$ , although there is some evidence that a finite  $\alpha$  (just as in an Ornstein-Zernike process) gives slightly better results, cf. Burgett (1989), Ch.6.

We are now ready to compute the functional  $S_2^1$ , eq. (3.2.11). Clearly, the first term reads:

$$S[U] = \int d^3x d^3x' n^i[U(\mathbf{x})] K_{ij}^T(\mathbf{x}, \mathbf{x}') n^j[U(\mathbf{x}')] \quad (4.8)$$

On expressing the average velocity as the curl of its vector potential,  $\mathbf{U} = \nabla \times \mathbf{A}$  and using (4.6), we obtain in the axial gauge:

$$S = a \int d^2x dz_1 dz_2 e^{-\alpha|z_1-z_2|} \left[ -\epsilon^{AS} \partial_3 A_S \epsilon^{RM} \partial_A \partial_R A_M \right. \\ \left. + \frac{1}{2} \partial_3 (\epsilon^{RS} \partial_R A_S)^2 \right]_{z_1} [ \dots ]_{z_2}, \quad (4.9)$$

where the second factor in square brackets has the same structure as the first one, but it is evaluated at  $z_2$ . We now notice that the second term in both square brackets in the last equation is a total derivative. Consequently, upon integration by parts, its contribution becomes proportional to  $\alpha$  and hence it is small for small values of that parameter. Therefore, that term can be omitted without substantially affecting the results and we shall do so in this paper.

We next compute the functional derivatives of (4.9) in order to generate the second term of  $S_2^1$  in eq. (3.2.11). This is a straight forward procedure and we merely quote the result.

$$\begin{aligned} \frac{\delta^2 S}{\delta A_A(x_1, z_1) \delta A_B(x_2, z_2)} = & 2a e^{-\alpha|z_1 - z_2|} \left\{ 2\alpha \epsilon(z_1 - z_2) \epsilon^{NA} \epsilon^{RB} \partial_R \partial_N \delta^2(x_1 - x_2) \right. \\ & \left. + \alpha^2 \epsilon^{NA} \epsilon^{RM} \partial_N \partial_R A_M \epsilon^{PB} \epsilon^{QS} \partial_P \partial_Q A_S \delta^2(x_1 - x_2) \right\}. \end{aligned} \quad (4.10).$$

In this equation and from now on,  $x_1$ , etc. stand for the transverse components of the position vector, whereas  $z_1$ , etc. denote the longitudinal component of the same vector. The symbol  $\epsilon(z)$  stands for the sign function,  $\epsilon(z) = z/|z|$ . Unlike in the expression of  $S$ , we now cannot omit terms of  $O(\alpha)$  or  $O(\alpha^2)$ , for there are no terms of  $O(1)$  present in eq. (4.10). However, we notice that the quantity  $\alpha$  can be scaled out of the expression  $S_2^1$ . In fact, on looking at the structure of eqs. (3.1.4), (3.2.11), (4.9) and (4.10), we realize that upon the rescaling,  $z \rightarrow (1/\alpha)z$ ,  $A \rightarrow (1/\alpha^{\frac{1}{2}})A$ , and  $Z_{AB} \rightarrow (1/\alpha^3)Z_{AB}$ , the expression of  $S_2^1$  is multiplied by an overall factor  $1/\alpha^4$ . This, however, can be absorbed into the constant entering the expression of the force correlation function; at any rate, the location of the stationary point of the functional  $S_2^1$  is independent of the value of  $\alpha$ . (We remark, however, that higher order approximations to  $S_2$  do not have this scaling property.)

The reader certainly notices that our approach has been quite general and the results obtained so far are applicable to almost any flow geometry. We now specialize to the case of a cylindrically symmetric free jet. Such jets have a characteristic velocity, namely the mean flow velocity at the centerline of the jet exit,  $U_{0m}$ , and a characteristic length,  $d$ , namely the diameter of the jet at the exit. From now on, as is customary, we measure all velocities and distances in these units. However, in order to keep the notation simple, this is not indicated explicitly in the formulae. (For instance, a velocity component denoted by  $U$  is understood to mean  $U/U_{0m}$ , or an axial distance,  $z$ , is to be interpreted as  $z/d$ .)

A cylindrically symmetric mean flow can be described in the axial gauge by means of a vector potential which has a tangential component only. Specifically, in cylindrical coordinates ( $x^1 = r \cos\varphi$ ,  $x^2 = r \sin\varphi$ ), we choose a trial function for the vector potential of

the form:

$$A_{\varphi} = \frac{1}{2} U(0,z) R(z)^2 [1 - \exp(-r^2/R(z)^2)], \quad (4.11)$$

with the other two components vanishing. Here  $U(0,z)$  stands for the mean velocity along the jet axis; the quantity  $R(z)$  characterizes the radius of the jet at distance  $z$  from the exit. The choice of such a functional form is motivated by its simplicity and by various fits and approximate calculations of free jets; see, e.g. the classic text of Hinze (1975). This vector potential gives rise to an axial and a radial component of the mean velocity of the form:

$$\begin{aligned} U_z(r,z) &= U(0,z) e^{-r^2/R(z)^2}, \\ U_r(r,z) &= \left\{ \frac{d \ln U(0,z)}{dz} \frac{1}{2} R(z)^2 (1 - e^{-r^2/R(z)^2}) \right. \\ &\quad \left. + R(z) \frac{dR}{dz} \left[ 1 - \left( 1 + \frac{r^2}{R^2} \right) e^{-r^2/R(z)^2} \right] \right\}. \end{aligned} \quad (4.12)$$

Next we impose the requirement of cylindrical symmetry on the two-point correlation function,  $Z_{AB}$ , cf. eq. (3.1.4):

$$Z_{AB} = \delta_{AB} Z_1 + \epsilon^{AB} Z_2. \quad (4.13)$$

In order to simplify the computation, we arbitrarily set  $Z_2 = 0$ . There is no strong physical motivation for this choice; it simplifies the calculations to a considerable extent, albeit at the cost of some loss of accuracy. We assume the following functional form for  $Z_1$ :

$$\begin{aligned} Z_1 &= A e^{-f(z_1 - z_2)^2 - g(x_1^A - x_2^A)(x_1^A - x_2^A)} \\ &\quad \times [U(0,z_1)U(0,z_2)]^d (1 + BM) e^{-\delta M}. \end{aligned} \quad (4.14)$$

In this equation, the parameters  $A$ ,  $B$ ,  $d$ ,  $\delta$ ,  $f$ , and  $g$  are treated as variational parameters,

although later we found that the results are rather insensitive to the choice of  $d$  and we ended up using a plausible value,  $d = 1/2$ . The expression  $M$  is defined as follows:

$$M = \frac{1}{2} \left( \frac{A}{R(z_1)} \frac{A}{x_1} + \frac{A}{R(z_2)} \frac{A}{x_2} \right). \quad (4.15)$$

The radius of the jet was assumed to be a linear function of  $z$ ,  $R(z) = a + bz$ , with the parameters  $a$  and  $b$  being also varied. We did not, however, vary the functional form of the mean velocity on the jet axis: although this is possible in principle, we used a fixed form in order to simplify the calculational task. We used the following simple modification of Spalding's formula (Hinze 1975), which is quite accurate for  $z \geq 2$ :

$$U(0,z) = \frac{1.35}{1 + 0.038z^{3/2}}. \quad (4.16)$$

The next task is to insert the expressions of the vector potential of the average flow and of the two point correlation function into the functional  $S_2^1$  and to minimize that expression with respect to the parameters. The operations are elementary, but extremely tedious to carry out by hand, although this is not impossible. They are, however, easily carried out by a symbolic manipulation program. We carried out the differentiations and a part of the integrations by using the program "Maple", (Char et al. 1985). The search for an extremum of the functional was done numerically. All computations were done on a Sun 3/160 computer. In order to carry out the search for the extremum, one needs starting values of the parameters. These were obtained by comparing  $U_z(r,z)$ , eq. (4.12), to a few data points from Zoltani and Bicen (1990a); we also chose, arbitrarily,  $f = g = \delta = 1$  as starting values. (The values of  $f$  and  $g$  cannot be read off directly from the data.)

The search resulted in the following values of the parameters:

$$\begin{aligned} a &= 0.25, \quad b = 0.076; \\ A &= 0.0013, \quad B = 2.3 \\ \delta &= 1.79, \\ f &\approx 5.6, \quad g \approx 3.7 \end{aligned} \quad (4.17)$$

In the approximation used, the value of the functional  $S_2$  changes rather slowly with variations of the parameters  $f$  and  $g$ . This is due to the fact that in this approximation, the interaction between the fluctuations and the mean flow is taken into account, but the mutual interaction of the fluctuations is neglected. (The functional  $S_2^1$  is almost local: it contains  $\delta$  functions and derivatives of  $\delta$  functions of finite order.) Hence, the values of  $f$  and  $g$  cannot be determined efficiently. This is not a serious deficiency, however: most measurements determine correlation functions at coincident arguments, which are rather insensitive to these parameters.

The correlation functions of velocity fluctuations are determined by inserting the expression of  $Z$  into eq. (3.1.5). We summarize the results by listing the expressions of the axial component of the mean flow and of the correlation functions at coincident arguments.

$$\begin{aligned}
G_{11} = G_{22} &= 2fA U_z(0,z) e^{-\chi^2} \{1 + B\chi^2 + (1/2f)[d(z)^2 \\
&\quad + \chi^2 B d(z) (d(z) - (B-1)/B b/R(z)) \\
&\quad + \chi^4 B b/R (d(z) + (1-2B)/4B b/R(z)) \\
&\quad + \chi^5 B/4 (b/R(z))^2\}, \\
G_{33} &= 4gA U_z(0,z) e^{-\chi^2} \{1 + \chi^2[B - (2B-1)/4R(z)^2g] \\
&\quad + \chi^4 B/4R(z)^2g\}, \\
G_{13} = G_{23} &= A \chi e^{-\chi^2} U(0,z) \{ (1-B) d(z) \\
&\quad + \chi^2 [B d(z) + (1-2B) b/R] + \chi^4 bB/R(z)\}, \\
U_z(r,z) &= U(0,z) e^{-\chi^2}. \tag{4.18}
\end{aligned}$$

Here,  $\chi^2 = \delta r^2/R(z)^2$  and  $d(z) = d/dz(U(0,z))^{1/2}$ .

The results listed in eq. (4.18) are evaluated using the parameter values (4.17) and compared with data taken from Zoltani and Bicen (1990a), (1990b) in Figures 1 thru 4. (The experimental circumstances there were not exactly identical. In particular, the Reynolds numbers differ by approximately 30% at the jet exit. However, at large Reynolds numbers the profile of the mean flow and the correlation functions should be insensitive to the precise value of  $Re$ , cf. the discussion at the beginning of this Section. The data bear out this conclusion.) In the figures we use conventional notation. The correspondence

between the notation used in this paper (better suited for theoretical calculations) and the conventional one, as used in Zoltani and Bicen (1990a), (1990b) is the following. (All arguments are suppressed, so, for instance,  $G_{11}$  stands for  $G_{11}(\mathbf{x}, \mathbf{x})$ . This does not lead to any ambiguity.)

$$\begin{aligned}
 G_{33} &= \frac{\langle u^2 \rangle}{U_{0m}^2}, \\
 G_{11} &= \frac{\langle v^2 \rangle}{U_{0m}^2}, \\
 G_{22} &= \frac{\langle w^2 \rangle}{U_{0m}^2}, \\
 G_{13} &= \frac{\langle uv \rangle}{U_{0m}^2}, \\
 U_z &= U
 \end{aligned}
 \tag{4.19}$$

## 5. DISCUSSION

A look at Figures 1 thru 4 shows that overall, the simple trial functions used in our variational calculation, with the parameters determined by extremizing  $S_2$ , are in a reasonable agreement with the data. (The agreement with the data is better at large values of  $z$ . This is understandable on physical grounds: at larger values of  $z$ , the statistics of the flow is closer to the stationary ensemble we have been working with.) In particular, our assumption of cylindrically symmetric correlation functions appears to be well justified. This is expected on physical grounds: to a very good approximation, the mean flow is cylindrically symmetric. Although this does not necessary imply that the correlation functions themselves should possess the symmetry, one expects violations of this symmetry to occur on relatively short time scales. Likewise, the assumption about "transverse scaling", namely that the radial dependence of both the mean flow and the correlation functions occurs only in the scale invariant combination,  $\chi^2 = \delta r^2 / R(z)^2$  appears to be reasonably well satisfied.

A notable exception is the correlation function  $G_{13}$ , which is predicted to be much too small compared to the data. Various modifications of the form of the trial functions have

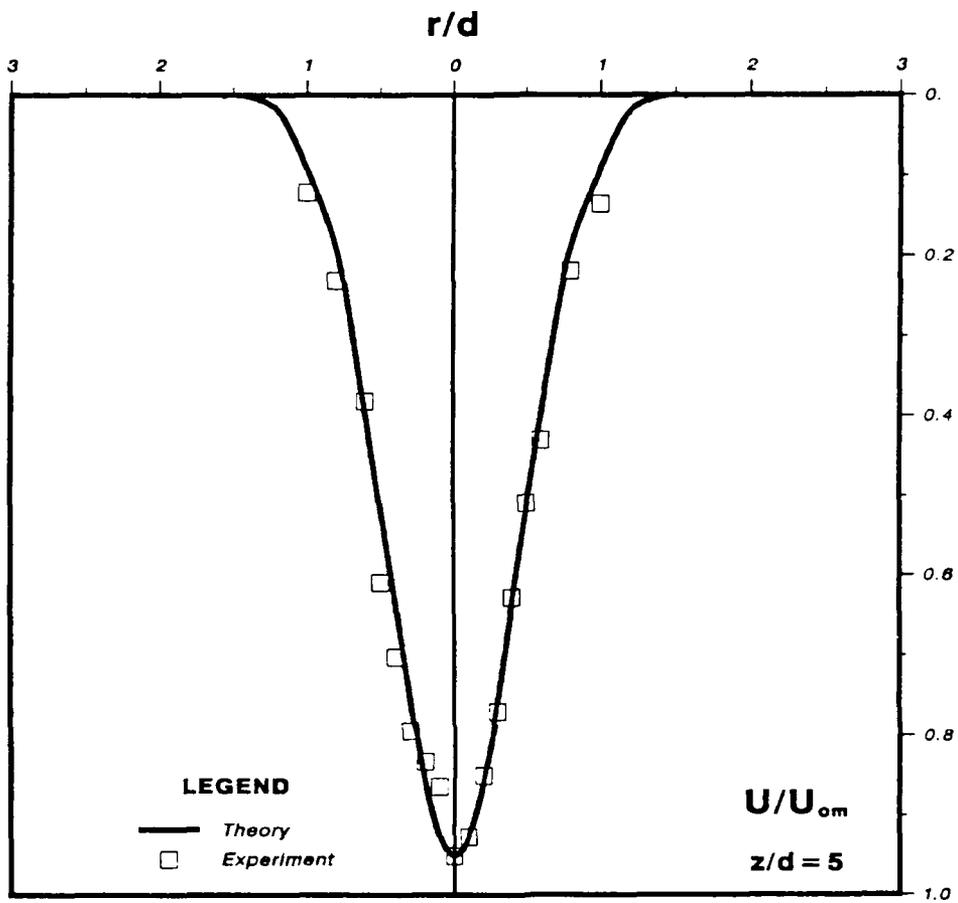


Figure 1. Predicted and Measured Mean Velocity Profiles at  $(z/d) = 5$ . The Values Were Normalized With the Centerline Jet Exit Velocity.

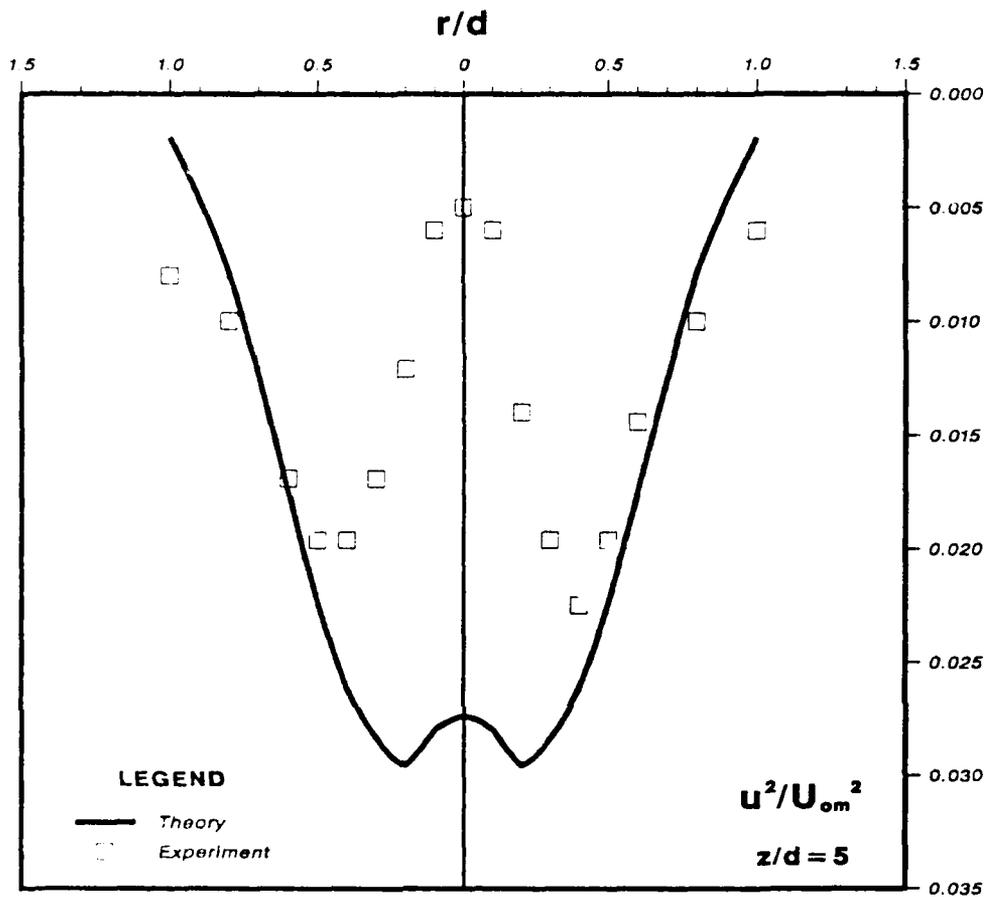


Figure 2. Predicted and Measured Values of the Autocorrelation Function,  $\overline{u^2}$  in the Axial Direction at  $(z/d) = 5$ . The Values Were Normalized With the Square of the Centerline Jet Exit Velocity.

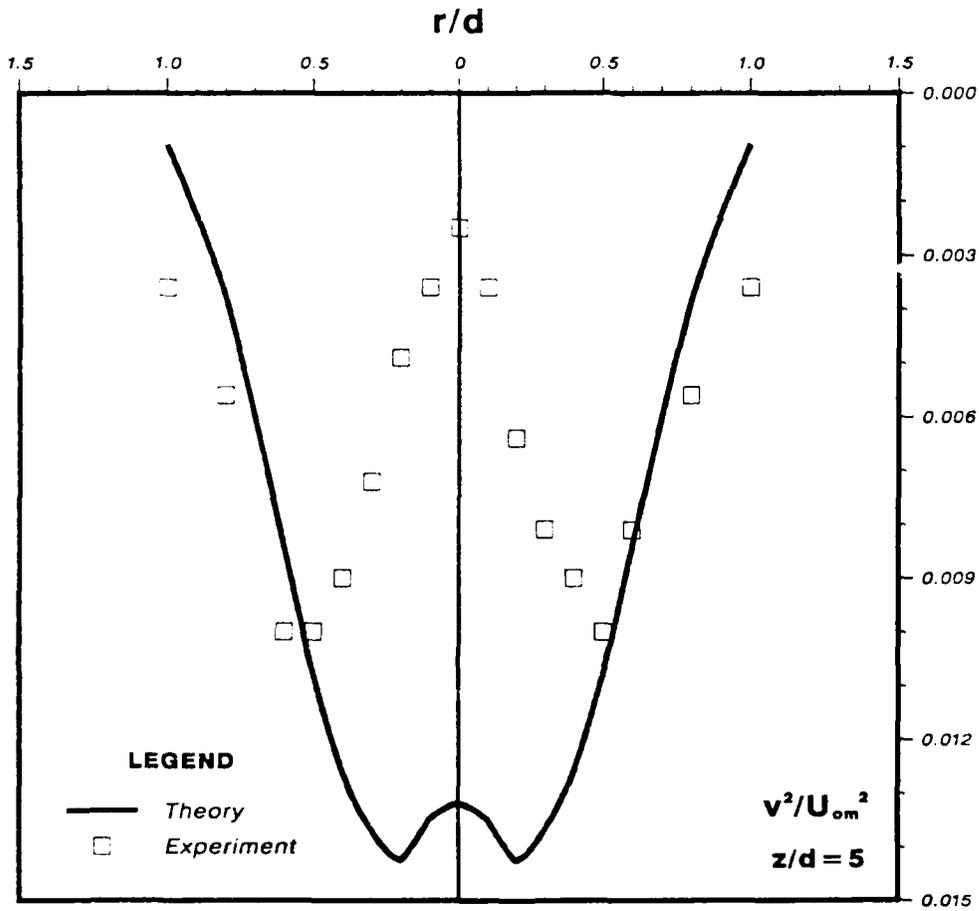


Figure 3. Predicted and Measured Autocorrelation Function,  $\langle v^2 \rangle$  in the Radial Direction at  $(z/d) = 5$ . The Values Were Normalized With the Square of the Centerline Jet Exit Velocity.

## REFERENCES

- Batchelor, G.K., "The Theory of Axisymmetric Turbulence", Proc. Roy. Soc. London, Vol. 186A, 480, 1946.
- Burgett, W.S., "The Calculation of Correlation Functions in the Statistical Theory of Turbulent Flows", PhD. diss., Johns Hopkins University. TIPAC Technical Report No. 8914, 1989.
- Chandrasekhar, S., "The Theory of Axisymmetric Turbulence", Phil. Trans. Roy. Soc. London, Vol. 242 A, 557, 1950.
- Char, B.W., K.O. Geddes, G.H. Gonnet and S.M. Watt, Maple Reference Manual. Waterloo, Ont.: University of Waterloo Computation Group, 1985.
- Cornwall J.M., R. Jackiw and E. Tomboulis, "Effective Action for Composite Operators", Phys. Rev. D, Vol. 10, 2428, 1974.
- De Dominicis, C. and P.C. Martin, "Stationary Entropy Principle and Renormalization in Normal and Superfluid Systems", J. Math. Phys. Vol. 5, 14, 1964.
- Domokos G., S. Kovesi-Domokos and C.K. Zoltani, "Random Systems, Turbulence and Disordering Fields", Physica A, Vol. 153, 84 (1988).
- Domokos D. and P. Suranyi, "Spontaneous Symmetry Breaking in Quantum Field Theory", Sov. J. Nuclear Phys., Vol. 2, 361, 1964.
- De Dominicis C. and L. Peliti, "Field Theory Renormalization and Critical Dynamics above  $T_c$ ", Phys. Rev. B, Vol. 18, 353(1978).
- Hinze, J.O., Turbulence. New York, NY: McGraw Hill, 1975. Chapter 6.
- Itzykson, C. and J.B. Zuber, Quantum Field Theory. New York, NY: McGraw Hill, 1980.
- Martin, P.C., E.D. Siggia and H.A. Rose, "Statistical Dynamics of Classical Systems",

Phys. Rev. A, Vol. 8, 423, (1973).

Trevino, G., "An Introduction to the Theory of Nonhomogenous Turbulence", Texas J. Sci.  
Vol. 34, 35, 1982.

Zoltani, C.K. and A.F. Bicen, "Velocity Measurements in a Turbulent, Dilute Two-Phase  
Jet", Experiments in Fluids, Vol. 9, 295-300 (1990a).

Zoltani, C.K. and A.F. Bicen "Effect of Initial Conditions on the Development of  
Two-Phase Jets", BRL-TR-3100, U.S. Army Ballistic Research Laboratory, Aberdeen  
Proving Ground, MD, April 1990b.

<u>No of</u> <u>Copies</u>	<u>Organization</u>	<u>No of</u> <u>Copies</u>	<u>Organization</u>
2	Administrator Defense Technical Info Center ATTN: DTIC-DDA Cameron Station Alexandria, VA 22304-6145	1	Commander US Army Missile Command ATTN: AMSMI-RD-CS-R (DOC) Redstone Arsenal, AL 35898-5010
1	HQDA (SARD-TR) WASH DC 20310-0001	1	Commander US Army Tank-Automotive Command ATTN: AMSTA-TSL (Technical Library) Warren, MI 48397-5000
1	Commander US Army Materiel Command ATTN: AMCDRA-ST 5001 Eisenhower Avenue Alexandria, VA 22333-0001	1	Director US Army TRADOC Analysis Command ATTN: ATRC-WSR White Sands Missile Range, NM 88002-5502
1	Commander US Army Laboratory Command ATTN: AMSLC-DL Adelphi, MD 20783-1145	(Class. only) 1	Commandant US Army Infantry School ATTN: ATSH-CD (Security Mgr.) Fort Benning, GA 31905-5660
2	Commander US Army, ARDEC ATTN: SMCAR-IMI-I Picatinny Arsenal, NJ 07806-5000	(Unclass. only) 1	Commandant US Army Infantry School ATTN: ATSH-CD-CSO-OR Fort Benning, GA 31905-5660
2	Commander US Army, ARDEC ATTN: SMCAR-TDC Picatinny Arsenal, NJ 07806-5000	1	Air Force Armament Laboratory ATTN: AFATL/DLODL Eglin AFB, FL 32542-5000
1	Director Benet Weapons Laboratory US Army, ARDEC ATTN: SMCAR-CCB-TL Watervliet, NY 12189-4050		<u>Aberdeen Proving Ground</u>
1	Commander US Army Armament, Munitions and Chemical Command ATTN: SMCAR-ESP-L Rock Island, IL 61299-5000	2	Dir, USAMSAA ATTN: AMXSY-D AMXSY-MP, H. Cohen
1	Director US Army Aviation Research and Technology Activity ATTN: SAVRT-R (Library) M/S 219-3 Ames Research Center Moffett Field, CA 94035-1000	1	Cdr, USATECOM ATTN: AMSTE-TD
		3	Cdr, CRDEC, AMCCOM ATTN: SMCCR-RSP-A SMCCR-MU SMCCR-MSI
		1	Dir, VLAMO ATTN: AMSLC-VL-D

<u>No. of</u> <u>Copies</u>	<u>Organization</u>
1	Commander USA Concepts Analysis Agency ATTN: D. Hardison 8120 Woodmont Avenue Bethesda, MD 20014-2797
1	C.I.A. 01R/DB/Standard Washington, DC 20505
1	US Army Ballistic Missile Defense Systems Command Advanced Technology Center P.O. Box 1500 Huntsville, AL 35807-3801
1	Chairman DoD Explosives Safety Board Room 85C-C Hoffman Bldg. 1 2461 Eisenhower Avenue Alexandria, VA 22331-0600
1	Commander US Army Materiel Command ATTN: AMCPM-GCM-WF 5001 Eisenhower Avenue Alexandria, VA 22333-5001
1	Commander US Army Materiel Command ATTN: AMCDE-DW 5001 Eisenhower Avenue Alexandria, VA 22333-5001
3	Project Manager Autonomous Precision-Guided Munition (APGM) US Army, ARDEC ATTN: AMCPM-CW AMCPM-CWW AMCPM-CWA-S, R. DeKleine Picatinny Arsenal, NJ 07806-5000
2	Project Manager Production Base Modernization Agency ATTN: AMSMC-PBM, A. Siklosi AMSMC-PBM-E, L. Laibson Picatinny Arsenal, NJ 07806-5000

<u>No. of</u> <u>Copies</u>	<u>Organization</u>
3	PEO-Armaments Project Manger Tank Main Armament Systems ATTN: AMCPM-TMA, K. Russell AMCPM-TMA-105 AMCPM-TMA-120, C. Roller Picatinny Arsenal, NJ 07806-5000
1	Commander US Army, ARDEC ATTN: SMCAR-AEE Picatinny Arsenal, NJ 07806-5000
13	Commander US Army, ARDEC ATTN: SMCAR-AEE-B, A. Beardell S. Bernstein P. Bostonian B. Brodman R. Cirincione D. Downs S. Einstein A. Grabowsky P. Hui J. O'Reilly N. Ross J. Rutkowski S. Westley Picatinny Arsenal, NJ 07806-5000
2	Commander US Army, ARDEC ATTN: SMCAR-AES, S. Kaplowitz D. Spring Picatinny Arsenal, NJ 07806-5000
2	Commander US Army, ARDEC ATTN: SMCAR-HFM, E. Barriores SMCAR-CCH-V, C. Mandala Picatinny Arsenal, NJ 07806-5000
1	Commander US Army, ARDEC ATTN: SMCAR-FSA-T, M. Salsbury Picatinny Arsenal, NJ 07806-5000
1	Commander, USACECOM R&D Technical Library ATTN: ASQNC-ELC-I-T, Myer Center Fort Monmouth, NJ 07703-5301

<u>No. of Copies</u>	<u>Organization</u>	<u>No. of Copies</u>	<u>Organization</u>
1	Commander US Army Harry Diamond Laboratories ATTN: SLCHD-TA-L 2800 Powder Mill Rd Adelphi, MD 20783-1145	1	Commander US Army Research Office ATTN: Technical Library P.O. Box 12211 Research Triangle Park, NC 27709-2211
1	Commandant US Army Aviation School ATTN: Aviation Agency Fort Rucker, AL 36360	1	Commander US Army Belvoir Research and Development Center ATTN: STRBE-WC Fort Belvoir, VA 22060-5006
1	Project Manager US Army Tank-Automotive Command Improved TOW Vehicle ATTN: AMCPM-ITV Warren, MI 48397-5000	1	Director US Army TRAC-Ft. Lee ATTN: ATRC-L, Mr. Cameron Fort Lee, VA 23801-6140
2	Program Manager US Army Tank-Automotive Command ATTN: AMCPM-ABMS, T. Dean (2 cys) Warren, MI 48092-2498	1	Commandant US Army Command and General Staff College Fort Leavenworth, KS 66027
1	Project Manager US Army Tank-Automotive Command Fighting Vehicle Systems ATTN: AMCPM-BFVS Warren, MI 48092-2498	1	Commandant US Army Special Warfare School ATTN: Rev and Trng Lit Div Fort Bragg, NC 28307
1	President US Army Armor and Engineer Board ATTN: ATZK-AD-S Fort Knox, KY 40121-5200	3	Commander Radford Army Ammunition Plant ATTN: SMCAR-QA/HI LIB (3 cys) Radford, VA 24141-0298
1	Project Manager US Army Tank-Automotive Command M-60 Tank Development ATTN: AMCPM-ABMS Warren, MI 48092-2498	1	Commander US Army Foreign Science and Technology Center ATTN: AMXST-MC-3 220 Seventh Street, NE Charlottesville, VA 22901-5396
1	Director HQ, TRAC RPD ATTN: ATCD-MA Fort Monroe, VA 23651-5143	2	Commander Naval Sea Systems Command ATTN: SEA 62R SEA 64 Washington, DC 20362-5101
2	Director US Army Materials Technology Laboratory ATTN: SLCMT-ATL (2 cys) Watertown, MA 02172-0001	1	Commander Naval Air Systems Command ATTN: AIR-954-Technical Library Washington, DC 20360

<u>No. of</u> <u>Copies</u>	<u>Organization</u>	<u>No. of</u> <u>Copies</u>	<u>Organization</u>
1	Assistant Secretary of the Navy (R, E, and S) ATTN: R. Reichenbach Room 5E787 Pentagon Bldg Washington, DC 20375	5	Commander Naval Surface Warfare Center ATTN: Code G33, J. L. East W. Burrell J. Joindrow Code G23, D. McClure Code DX-21 Technical Library Dahlgren, VA 22448-5000
1	Naval Research Laboratory Technical Library Washington, DC 20375	3	Commander Naval Weapons Center ATTN: Code 388, C. F. Price Code 3895, T. Parr Information Science Division China Lake, CA 93555-6001
1	Commandant US Army Command and General Staff College Fort Leavenworth, KS 66027	1	OSD/SDIO/IST ATTN: Dr. H. Caveny Pentagon Washington, DC 20301-7100
2	Commandant US Army Field Artillery Center and School ATTN: ATSF-CO-MW, E. Dublisky (2 cys) Fort Sill, OK 73503-5600	3	Commander Naval Ordnance Station ATTN: T. C. Smith D. Brooks Technical Library Indian Head, MD 20640-5000
1	Office of Naval Research ATTN: Code 473, R. S. Miller 800 N. Quincy Street Arlington, VA 22217-9999	1	AL/TSTL (Technical Library) ATTN: J. Lamb Edwards AFB, CA 93523-5000
3	Commandant US Army Armor School ATTN: ATZK-CD-MS, M. Falkovitch (3 cys) Armor Agency Fort Knox, KY 40121-5215	1	AFATL/DLYV Eglin AFB, FL 32542-5000
2	Commander US Naval Surface Warfare Center ATTN: J. P. Consaga C. Gotzmer Indian Head, MD 20640-5000	1	AFATL/DLXP Eglin AFB, FL 32542-5000
4	Commander Naval Surface Warfare Center ATTN: Code 240, S. Jacobs Code 730 Code R-13, K. Kim R. Bernecker Silver Spring, MD 20903-5000	1	AFATL/DLJE Eglin AFB, FL 32542-5000
2	Commanding Officer Naval Underwater Systems Center ATTN: Code 5B331, R. S. Lazar Technical Library Newport, RI 02840	1	NASA/Lyndon B. Johnson Space Center ATTN: NHS-22 Library Section Houston, TX 77054
		1	AFELM, The Rand Corporation ATTN: Library D 1700 Main Street Santa Monica, CA 90401-3297

<u>No. of Copies</u>	<u>Organization</u>	<u>No. of Copies</u>	<u>Organization</u>
3	AAI Corporation ATTN: J. Herbert J. Frankle D. Cleveland P.O. Box 126 Hunt Valley, MD 21030-0126	3	Lawrence Livermore National Laboratory ATTN: L-355, A. Buckingham M. Finger L-324, M. Constantino P.O. Box 808 Livermore, CA 94550-0622
2	Aerojet Solid Propulsion Company ATTN: P. Micheli L. Torreyson Sacramento, CA 96813	1	Olin Corporation Badger Army Ammunition Plant ATTN: F. E. Wolf Baraboo, WI 53913
1	Atlantic Research Corporation ATTN: M. King 5390 Cherokee Avenue Alexandria, VA 22312-2302	2	Olin Corporation Smokeless Powder Operation ATTN: E. J. Kirschke A. F. Gonzalez P.O. Box 222 St. Marks, FL 32355-0222
3	AL/LSCF ATTN: J. Levine L. Quinn T. Edwards Edwards AFB, CA 93523-5000	1	Paul Gough Associates, Inc. ATTN: Dr. Paul S. Gough 1048 South Street Portsmouth, NH 03801-5423
1	AVCO Everett Research Laboratory ATTN: D. Stickler 2385 Revere Beach Parkway Everett, MA 02149-5936	1	Physics International Company ATTN: Library, H. Wayne Wampler 2700 Merced Street San Leandro, CA 98457-5602
2	Calspan Corporation ATTN: C. Murphy (2 cys) P.O. Box 400 Buffalo, NY 14225-0400	1	Princeton Combustion Research Laboratory, Inc. ATTN: M. Summerfield 475 US Highway One Monmouth Junction, NJ 08852-9650
1	General Electric Company Armament Systems Department ATTN: J. Mandzy 128 Lakeside Avenue Burlington, VT 05401-4985	2	Rockwell International Rocketdyne Division ATTN: BA08, J.E. Flanagan J. Gray 6633 Canoga Avenue Canoga Park, CA 91303-2703
1	IITRI ATTN: M. J. Klein 10 W. 35th Street Chicago, IL 60616-3799	3	Thiokol Corporation Huntsville Division ATTN: D. Flanigan Dr. John Deur Technical Library Huntsville, AL 35807
1	Hercules, Inc. Allegheny Ballistics Laboratory ATTN: William B. Walkup P.O. Box 210 Rocket Center, WV 26726		
1	Hercules, Inc. Radford Army Ammunition Plant ATTN: E. Hibshman Radford, VA 24141-0299		

<u>No. of</u> <u>Copies</u>	<u>Organization</u>
2	Thiokol Corporation Elkton Division ATTN: R. Biddle Technical Library P.O. Box 241 Elkton, MD 21921-0241
1	Veritay Technology, Inc. ATTN: E. Fisher 4845 Millersport Highway East Amherst, NY 14501-0305
1	Universal Propulsion Company ATTN: H. J. McSpadden Black Canyon Stage 1 Box 1140 Phoenix, AZ 84029
1	Battelle Memorial Institute ATTN: Technical Library 505 King Avenue Columbus, OH 43201-2693
1	Brigham Young University Department of Chemical Engineering ATTN: M. Beckstead Provo, UT 84601
1	California Institute of Technology 204 Karman Laboratory Main Stop 301-46 ATTN: F.E.C. Culick 1201 E. California Street Pasadena, CA 91109
1	California Institute of Technology Jet Propulsion Laboratory ATTN: L. D. Strand, MS 512/102 4800 Oak Grove Drive Pasadena, CA 91109-8099
1	University of Illinois Department of Mechanical/Industrial Engineering ATTN: H. Krier 144 MEB; 1206 N. Green Street Urbana, IL 61801-2978
1	University of Massachusetts Department of Mechanical Engineering ATTN: K. Jakus Amherst, MA 01002-0014

<u>No. of</u> <u>Copies</u>	<u>Organization</u>
1	University of Minnesota Department of Mechanical Engineering ATTN: E. Fletcher Minneapolis, MN 55414-3368
3	Georgia Institute of Technology School of Aerospace Engineering ATTN: B.T. Zim E. Price W.C. Strahle Atlanta, GA 30332
1	Institute of Gas Technology ATTN: D. Gidaspow 3424 S. State Street Chicago, IL 60616-3896
1	Johns Hopkins University Applied Physics Laboratory Chemical Propulsion Information Agency ATTN: T. Christian Johns Hopkins Road Laurel, MD 20707-0690
1	Massachusetts Institute of Technology Department of Mechanical Engineering ATTN: T. Toong 77 Massachusetts Avenue Cambridge, MA 02139-4307
1	Pennsylvania State University Applied Research Laboratory ATTN: G. M. Faeth University Park, PA 16802-7501
1	Pennsylvania State University Department of Mechanical Engineering ATTN: K. Kuo University Park, PA 16802-7501
1	Purdue University School of Mechanical Engineering ATTN: J. R. Osborn TSPC Chaffee Hall West Lafayette, IN 47907-1199
1	SRI International Propulsion Sciences Division ATTN: Technical Library 333 Ravenwood Avenue Menlo Park, CA 94025-3493

<u>No. of Copies</u>	<u>Organization</u>	<u>No. of Copies</u>	<u>Organization</u>
1	Rensselaer Polytechnic Institute Department of Mathematics Troy, NY 12181	1	Alliant Techsystems, Inc. ATTN: R. E. Tompkins MN38-3300 5700 Smetana Drive Minnetonka, MN 55343
2	Director Los Alamos Scientific Laboratory ATTN: T3, D. Butler M. Division, B. Craig P.O. Box 1663 Los Alamos, NM 87544	1	Science Applications, Inc. ATTN: R. B. Edelman 23146 Cumorah Crest Drive Woodland Hills, CA 91364-3710
1	General Applied Sciences Laboratory ATTN: J. Erdos 77 Raynor Avenue Ronkonkama, NY 11779-6649	1	Battelle Columbus Laboratories ATTN: Mr. Victor Levin 505 King Avenue Columbus, OH 43201-2693
1	Battelle PNL ATTN: Mr. Mark Garnich P.O. Box 999 Richland, WA 99352	1	Allegheny Ballistics Laboratory Propulsion Technology Department Hercules Aerospace Company ATTN: Mr. Thomas F. Farabaugh P.O. Box 210 Rocket Center, WV 26726
1	Stevens Institute of Technology Davidson Laboratory ATTN: R. McAlevy, III Castle Point Station Hoboken, NJ 07030-5907	1	MBR Research Inc. ATTN: Dr. Moshe Ben-Reuven 601 Ewing St., Suite C-22 Princeton, NJ 08540
1	Rutgers University Department of Mechanical and Aerospace Engineering ATTN: S. Temkin University Heights Campus New Brunswick, NJ 08903		<u>Aberdeen Proving Ground</u>  Cdr, CSTA ATTN: STECS-LI, R. Hendricksen
1	University of Southern California Mechanical Engineering Department ATTN: OHE200, M. Gerstein Los Angeles, CA 90089-5199		
2	University of Utah Department of Chemical Engineering ATTN: A. Baer G. Flandro Salt Lake City, UT 84112-1194		
1	Washington State University Department of Mechanical Engineering ATTN: C. T. Crowe Pullman, WA 99163-5201		

INTENTIONALLY LEFT BLANK.

## USER EVALUATION SHEET/CHANGE OF ADDRESS

This Laboratory undertakes a continuing effort to improve the quality of the reports it publishes. Your comments/answers to the items/questions below will aid us in our efforts.

1. BRL Report Number BRL-TR-3192 Date of Report JANUARY 1991
2. Date Report Received \_\_\_\_\_
3. Does this report satisfy a need? (Comment on purpose, related project, or other area of interest for which the report will be used.) \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_
4. Specifically, how is the report being used? (Information source, design data, procedure, source of ideas, etc.) \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_
5. Has the information in this report led to any quantitative savings as far as man-hours or dollars saved, operating costs avoided, or efficiencies achieved, etc? If so, please elaborate. \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_
6. General Comments. What do you think should be changed to improve future reports? (Indicate changes to organization, technical content, format, etc.) \_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_  
\_\_\_\_\_

**CURRENT ADDRESS**

Name \_\_\_\_\_

Organization \_\_\_\_\_

Address \_\_\_\_\_

City, State, Zip Code \_\_\_\_\_

7. If indicating a Change of Address or Address Correction, please provide the New or Correct Address in Block 6 above and the Old or Incorrect address below.

**OLD ADDRESS**

Name \_\_\_\_\_

Organization \_\_\_\_\_

Address \_\_\_\_\_

City, State, Zip Code \_\_\_\_\_

-----FOLD HERE-----

**DEPARTMENT OF THE ARMY**

Director  
U.S. Army Ballistic Research Laboratory  
ATTN: SLCBR-DD-T  
Aberdeen Proving Ground, MD 21005-5066  
**OFFICIAL BUSINESS**



**NO POSTAGE  
NECESSARY  
IF MAILED  
IN THE  
UNITED STATES**

**BUSINESS REPLY MAIL**  
FIRST CLASS PERMIT No 0001, APG, MD

POSTAGE WILL BE PAID BY ADDRESSEE

Director  
U.S. Army Ballistic Research Laboratory  
ATTN: SLCBR-DD-T  
Aberdeen Proving Ground, MD 21005-9989



-----FOLD HERE-----