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| COR: Daniel Weber, SMCUR-RS<br>FIELD GROUP SUB-GROUP<br>15 06 03<br>ABSTRACT (Continue on reverse if necessary<br>A numerical simulation of the<br>coaxial circular cylinders wa<br>actual behavior of a missile<br>simulation was performed in t<br>payload liquid, which compris<br>achieve full spin up relative<br>part, the despin moments acti<br>spun up liquid payload were c | BP-A, (301) 671-<br>18 SUBJECT TERMS (<br>Liquid Pay<br>Computatio<br>Finite Ele<br>and identify by block of<br>the flow of a liquid<br>with a liquid f<br>two parts. The<br>sed both Newton<br>to the missil-<br>ing on the spin<br>determined from<br>reference fram-<br>ckage FIDAP. T | -2186<br>Continue on reverse<br>yloads<br>onal Fluid D<br>ements<br>uid in the a<br>This simulat<br>payload duri<br>first part<br>ian and non-<br>e during lau<br>ning and nut<br>the steady-<br>e. The simu<br>he majority | e if necessary a<br>hynamics N<br>nnular reg<br>ion corres<br>ng its lau<br>was to det<br>Newtonian<br>nch. Duri<br>ating miss<br>state Navi<br>lations we<br>of the com | nd identify by t<br>ewtonian F<br>un-Newtoni<br>ponds to t<br>nch phase.<br>ermine if<br>fluids, wo<br>ng the sec<br>ile with a<br>er-Stokes<br>re perform<br>putations | block number)<br>luia<br>an Fluid<br>che<br>The<br>the<br>buld<br>cond<br>fully<br>equations<br>were done |
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#### PREFACE

The work described in this report was authorized under Contract No. DAAD05-86-M-Q985. This work was started in September 1986 and completed in September 1987.

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SIMULATIONS OF A MISSILE WITH LIQUID PAYLOAD UNDER LAUNCH CONDITIONS

### INTRODUCTION

This project was concerned with the numerical simulation of the flow of a liquid in the annular region between two coaxial circular cylinders. The investigations related to the behavior of a missile with a liquid payload during and immediately following launch.

The project comprised two Parts. In Part 1 solutions were computed of the transient Navier-Stokes equations for a number of different liquids, both Newtonian and non-Newtonian, with the object of determining whether or not the liquid became fully spun up over a given interval of time. In Part II the steady-state Navier-Stokes equations relative to an aeroballistic frame were solved with the aim of determining the despin moment on a spinning and nutating missile in a fully-spun-up mode.

Detailed descriptions of the two problems will be given below. In both Parts the simulations were performed with the finite element software package FIDAP. Most of the computations were done on an HP 9000 minicomputer; some were done on a Microvax workstation.

# Part J

# SPIN UP

# Formulation

Computations were performed to determine the spin up characteristics of a number of different liquids, both Newtonian and non-Newtonian, under conditions simulating the launch of a missile with a liquid payload. The essential features of the process studied were:

- the time interval over which spin up took place was from 0.01 sec to 0.12 sec;
- ii. the spin rate was accelerated over this time interval in a prescribed manner (detailed below) from zero to 11.8 rev/sec.
- iii. the axial acceleration varied over the same time interval in a prescribed manner (detailed below), with a value of 585 m/sec<sup>2</sup> at the end of the interval.

The liquid was taken to occupy the region between two coaxial circular cylinders of radii

$$R_1 = 2.625"$$
,  $R_2 = 4.5"$ 

respectively. The boundaries of the annular region were rigid and the length of the cylinders was 66". However, the liquid filled only 70% of the region, so that the effective (mean) length of the liquid column was

The equations governing the motion of the liquid are the Navier-Stokes equations and the continuity equation, which can be written as

$$\rho(\underline{v}^{*}\underline{v}^{*} + \underline{v}^{*} \cdot \nabla^{*}\underline{v}^{*}) = -\nabla^{*}\rho^{*} + \nabla^{*} \cdot \underline{T}^{*} + \rho \underline{F}^{*}$$
(1)  
$$\nabla^{*} \cdot \underline{v}^{*} = 0$$
(2)

$$\underline{T}^* = \mu_0 \underline{G}^* \quad (\underline{Y}^*) \tag{3}$$

where  $\mu_{o}$  is zero-shear-rate viscosity and where

$$\dot{\underline{\chi}}^* = \underline{h} (\nabla^* \underline{\chi}^* + \nabla^* \underline{\chi}^{*T}).$$
(4)

For Newtonian liquids (3) is just a linear relation; for the non-Newtonian liquids studied  $\underline{G}^*$  was given by empirical data.

## Nondimensionalization

It is convenient to express the governing equations and associated quantities in nondimensional form. All lengths are scaled with respect to the outer radius  $R_2$ , time with respect to the final angular velocity  $\omega$  (-11.8 rev/sec), and velocities with respect to the velocity  $R_2\omega$ . Thus we write

$$\underline{v}^* = R_2 \omega \underline{v}, \ t^* = \omega^{-1} t, \ \underline{r}^* = R_2 \underline{r}, \ p^* = \rho (R_2 \omega)^2 p,$$

$$\underline{T}^* = \mu_0 \omega \underline{T}, \ \underline{E}^* (t^*) = R_2 \omega^2 \underline{E} (t), \ \underline{y}^* = \omega \underline{y}.$$

Then the governing equations  $(1) \cdot (4)$  become

$$\mathbf{y}_{t} + \mathbf{y} \cdot \nabla \mathbf{y} = \cdots \nabla p + \frac{1}{Re} \nabla \cdot \underline{T} + E(t)$$
(5)

$$\nabla \cdot \underline{v} = 0 \tag{6}$$

with

$$\underline{T} = \underline{G}(\dot{\underline{X}}) \tag{7}$$

and

$$\underline{\gamma} = \overline{\gamma} (\nabla \underline{v} + \nabla \underline{v}^{T})$$
(8)

The parameter appearing in equation (5) is the Reynolds number based on zero-shear-rate viscosity,

$$Re = \frac{\mu \tilde{\kappa}_2^2 \omega}{\mu_o}$$
(9)

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We now specify in more detail certain scalings used in the calculations. The final angular velocity is

$$\omega = 11.8 \text{ revs/sec} = 74.14 \text{ rads/sec}$$

so that

The time of spin up is from  $t^* = 0.01$  sec to  $t^* = 0.12$  sec, and the spin rate increases linearly from  $t^* = 0.01$  sec to  $t^* = 0.08$  sec, and then is constant. We therefore define a dimensionless time t by the formula

$$t^* = .01 + .0135c$$

or, equivalently,

$$t = 7 + .14 (t^{-} - .01).$$

The relation between dimensional and dimensionless spin up times is therefore as shown in the following table

| t*   | t      |  |  |
|------|--------|--|--|
| . 01 | 0.0000 |  |  |
| . 08 | 5.1900 |  |  |
| .12  | 8,1554 |  |  |

Thus the total dimensionless spin up time is t = 8.1554.

We can also conveniently define a dimensionless angular velocity  $\Omega$  which changes in time, as indicated above, as follows:

$$\Omega = \Omega(z) = \begin{cases} t/5.19 & \text{for } 0 \le t \le 5.19 \\ 1 & \text{for } 5.19 \le t \le 8.1554 \end{cases}$$
(10)

It is convenient also to non-dimensionalize the time-dependent body force. This increases until  $t^* = 0.08$  and then experiences a small decrease. At the final instant  $t^* = 0.12$  it is given that

$$F_{\rm final}^* = 585.50 \,\,{\rm m/sec}^2$$

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When this is scaled with respect to  $R_2\omega^2$  we obtain

$$F_{\text{final}} = F_{\text{final}}^* / R_2 \omega^2 = 0.9319$$

Hence we write the dimensionless acceleration over the time interval of interest as

$$F(t) = -0.9319 f(t)$$
(11)

where

$$f(t) = \begin{cases} .1115 + .1790t & 0 \le t \le 5.19 \\ 1.1108 - 0.0136t & 5.19 \le t \le 8.1554 \end{cases}$$
(12)

The values at different times represented by (11) and (12) are indicated in the following table:

| t      | f(t)   | F(t) | ٤.*  | $F^{*}(t^{*})$ |
|--------|--------|------|------|----------------|
| 0      | .1115  | 1039 | .01  | - 65.28        |
| 5.19   | 1.0403 | 9695 | . 08 | - 609.1        |
| 8.1554 | 1.0000 | 9319 | .12  | - 585.5        |

The simulations were performed ton a tetal of 15 different liquids. Five were Newtonian and ten were non-Newtonian. For identification purposes

the ten non-Newtonian liquids were labeled 'iquid 1, liquid 2,...,liquid 10. The Newtonian liquids were labeled liquid 1N, liquid 2N, liquid 5N, liquid 6N and liquid 8N. This was because of equality between the viscosity of liquid 1N with the zero-shear-rate viscosity of liquid 1, etc. These viscosities and the corresponding Reynolds numbers are shown in Table 1. The dependence of the non-Newtonian liquids' viscosities on shear rate was determined from empirical data supplied.

| liquid<br># | μ <sub>0</sub><br>poise | Re    |
|-------------|-------------------------|-------|
| 1           | 1000                    | 9.688 |
| 1 <i>N</i>  | 1000                    | 9.688 |
| 2           | 600                     | 16.15 |
| 2N          | 600                     | 16.15 |
| 3           | .380                    | 25.50 |
| 4           | 200                     | 48.44 |
| 5           | 100                     | 96.88 |
| 5₽          | 100                     | 96.88 |
| 6           | 60                      | 161.5 |
| 6 <i>N</i>  | 60                      | 161.5 |
| 7           | 20                      | 484.4 |
| 8           | 10                      | 968.8 |
| 8 <i>N</i>  | 10                      | 968.8 |
| 9           | 6                       | 1615  |
| 10          | 2                       | 4844  |

TABLE 1. Liquids and zero-shear-rate viscosities

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The initial conditions for the simulation were that the liquid was at rest at (dimensionless) time t = 0.

The boundary conditions are applied to all the surfaces. The inner and outer curved boundaries are at dimensionless radial positions 0.5833 and 1.0 respectively. The lower, solid end boundary of the cylinder is at z = 0 in a cylindrical polar coordinate system, while the mean position of the upper free surface is at z = 10.27.

The boundary conditions are:

1. On the inner curved surface  $r \approx 0.5833$ , 0 < z < 10.27,

u - w - 0,  $v = 0.5833 \Omega(t)$ 

where  $\Omega(t)$  is given by equation (10); (u, v, w) are the velocity components in the radial, azimuthal and axial directions respectively.

ii. On the outer curved surface r = 1.0, 0 < z < 10.27,

u - w = 0,  $v = 1.\Omega(t)$ 

iii. On the base z = 0,

u - w - 0,  $v - r\Omega(t)$ , 0.5833 < r < 1.0.

IV. On the upper free surface it is assumed that the surface tension is zero. The boundary conditions on this surface are then that the normal velocity is zero and that the tangential stress is zero. These translate into

$$\mathbf{w} = \frac{\partial \mathbf{u}}{\partial \mathbf{z}} = \frac{\partial \mathbf{v}}{\partial z} = 0$$

on the mean position z = 10.27.

### Computations and Results

The problem was solved as a transient problem over the time interval

0 < t < 8.1554. Fixed time increments were used of length 0.02039, so that 40 time steps were required to complete the simulation. Each run took approximately 8 hours of c.p.u. time on an HP 9000 computer, except for liquids 9 and 10, where the run time became excessively long because of the need to design an extremely fine mesh.

The results are presented in the accompanying graphs. Figures 1-13 show the behavior of the azimuthal component of velocity at one particular point in the liquid as a function of time. This point was chosen (arbitrarily) to have a radial coordinate r = 0.756. The velocity at this point was zero at t = 0, and if the liquid is fully spun up into rigid body motion, it should have an azimuthal velocity v = 0.756 at the end of the time interval t = 8.155. The figures show that the Newtonian liquids 1N and 2N are indeed fully spun up, while none of the others is completely spun up. The non-Newtonian liquids 1 and 2 are nearly spun up, to within a few percent. The velocity at this point which is actually reached is a measure of the extent to which complete spin up has been achieved. Using the ratio of velocity attained to the spin up value (.756) as a simple measure, we can refer to percentages as shown in Table 2. Liquids 9 and 10 are so far from being spun up that the solutions were not calculated; they are in fact less than 1% spun up. Table 2 shows the dramatic decrease in the extent of spin up as viscosity decreases, as well as the significant effects of shear thinning.

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| liquid<br># | Percentage<br>spin up |
|-------------|-----------------------|
| 1           | 97                    |
| 1 <i>N</i>  | 100                   |
| 2           | 95                    |
| 2N          | 100                   |
| 3           | 89                    |
| 4           | 66                    |
| 5           | 40                    |
| 5 <i>N</i>  | 92                    |
| 6           | 26                    |
| 6N          | 79                    |
| 7           | 4                     |
| 8           | 1                     |
| 8 <i>N</i>  | 11                    |

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TABLE 2. Percentage spin up



Figure 1



Figure 2



Figure 3

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Figure 4



1. M. M. M.

Figure 5

(NA)

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Figure 7



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Figur. 10





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## DESPIN MOMENT

#### Formulation

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Computations were performed to determine the despin moment on the container described in Part I on the assumption that the liquid had become fully spun up, and that the system was moving under spin and nutation at a prescribed coning angle. The angular velocity of the spin is denoted  $\cdots$ and the angular velocity of the nutation is denoted  $\Omega$ . The axis of the cylinder is inclined to the vertical at the coning angle  $\theta$ . The two axes of rotation intersect at the midpoint of the base of the cylinder.

The problem was solved by formulating the governing equations in a nutating frame of reference, the so-called aeroballistic frame. In this coordinate system, the z-axis coincides with the axis of the cylinder, the x-axis lies in the plane containing the angular velocity vectors  $\underline{\omega}$  and  $\underline{\Omega}$ , and the y-axis is perpendicular to this frame. This constitutes a right-handed cartesian system.

The governing equations were written in dimensionless form, with lengths, velocity, pressure, stress, strain rate and body force made non-dimensional according to the same scalings as were used in Part I. The angular velocity vector  $\Omega$  of nutation is non-dimensionalized with respect to its magnitude  $\Omega$ . The governing equations for steady motion in the aeroballistic frame are then found to be

$$\underline{\mathbf{v}} \cdot \nabla \underline{\mathbf{v}} + 2\eta \underline{\Omega} \mathbf{x} \underline{\mathbf{v}} + \eta^2 \underline{\Omega} \mathbf{x} (\underline{\Omega} \mathbf{x} \underline{\mathbf{r}}) = -\nabla \mathbf{p} + \frac{1}{Re} \nabla \cdot \underline{\mathbf{T}} + \underline{F}$$
(13)

$$\nabla \cdot \underline{\mathbf{v}} = \mathbf{0} \tag{14}$$

where Re is the Reynolds number defined by equation (9) and  $\eta$  is the spin ratio

$$-\frac{\Omega}{\omega}$$
 (15)

The dimensionless angular velocity  $\Omega$  with respect to the cartesian aeroballistic frame is

ŋ

$$\underline{\Omega} = -\sigma \underline{i} + \sqrt{1 - \sigma^2} \underline{k} \tag{16}$$

where

$$\sigma = \sin \theta \tag{17}$$

is the third principal parameter of the problem.

The boundary conditions imposed for the solution of the problem were:

- i) the normal velocity component is zero at all rigid boundaries;
- the tangential velocity at rigid boundaries is equal to the rotational velocity of that boundary;
- iii) the normal velocity and the tangential stress are zero at the free surface of the liquid.

#### Computations and Results

Solutions were computed for six different cases: three of these related to Newtonian liquid 1N of Part 1, and three to non-Newtonian liquid 1. These liquids have (zero-shear-rate) viscosity of about 1000 poise and, as shown in Part 1, were both fully spun up. The various cases considered reflected the spin rate and coning rate at different instants during flight.

The quantity of interest was the despin moment denoted  $M_z$  in dimensionless form and  $M_z^*$  in dimensional form. The relation between the two, in accordance with the scaling of Part 1, is

$$M_z^* - \rho R_2^5 \omega^2 M_z$$

Once the velocity field has been computed using FIDAP, the despin moment is calculated with the aid of a simple subroutine.

# The results of the computations are shown in Table 3.

## Newtonian Liquid

| Case<br># | ω<br>Σpm | 0<br>rpm | η<br>spin<br>ratio | Re   | Ą       | M <sup>*</sup> <sub>z</sub><br>ft·lbs |
|-----------|----------|----------|--------------------|------|---------|---------------------------------------|
| 1         | 640      | 54       | . 0844             | 8.75 | .000324 | .0211                                 |
| 2         | 255      | 230      | . 9020             | 3.49 | .009453 | .0975                                 |
| 3         | 272      | 136      | . 5000             | 3.72 | .003188 | .0374                                 |

#### Non-Newtonian Liquid

| Case<br># | ω<br>rpm | Ω<br>rpm | η<br>spin<br>ratio | Re   | M       | M <sup>*</sup> _<br>ft-lbs |
|-----------|----------|----------|--------------------|------|---------|----------------------------|
| 1         | 640      | 54       | .0844              | 8.75 | .001384 | . 0899                     |
| 2         | 255      | 230      | . 9020             | 3.49 | .044821 | .4623                      |
| 3         | 272      | 136      | . 5000             | 3.72 | .011733 | . 1.377                    |

## TABLE 3. Despin Moments

It is clear that the non-Newtonian liquid exerts a despin moment about four times as great as the Newtonian liquid with the same zero-shearrate viscosity, all other parameters being equal. This can presumably be attributed to the shear thinning effects present in the non-Newtonian

liquid and the corresponding change in Reynolds number. However, as can be seen from Table 3, Reynolds number is not the only parameter contributing to variation in despin moment, so that the nonlinear modification of the velocity field in the non-Newtonian case must be regarded as a significant contributing factor to the outcome.

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