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SOFTWARE SUPPORTABILITY RISK ASSESSMENT IN OTAE  
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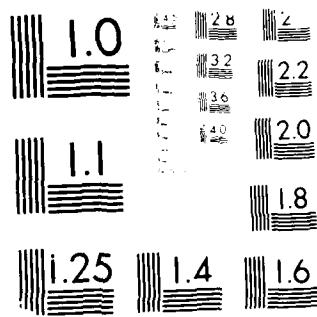
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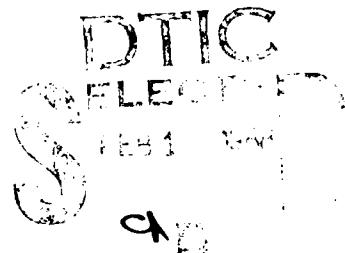


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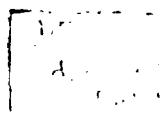
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# Software Supportability Risk Assessment in OT&E: Historical Baselines for Risk Profiles

Volume II



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October 7, 1985

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HISTORICAL BASELINES FOR RISK PROFILES  
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A-1

### **C. Data Survey Format**

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9. ABSTRACT *(Continue on reverse if necessary and identify by block numbers)*  
Assessing the software supportability risk of Air Force acquired systems is necessary to enable various decision makers to properly plan for system deployment. Risk assessment (RA) is required throughout the system acquisition life cycle. Since the perspective of OT&E is focused upon the overall system mission, including supportability, methods are required which point software testers to areas which require testing emphasis and which provide decision makers with an assessment of software and software support risk for production decisions. Due to the complexity of these requirements, it is necessary to develop and implement a risk assessment methodology of software supportability with the proper system mission perspective to ultimately assist the top level decision maker.

In the assessment of risk, the first criteria to establish are the baselines against which to measure the risk. This report contains the results of a study which collected software support activity data from a variety of DoD software support facilities and

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Item 11 (cont'd):

Historical Baselines for Risk Profiles (Volumes I and II)

Item 19 (cont'd):

systems. The data collected was used to develop historical profiles of the activities observed. These profiles are the risk baselines against which negative outcomes can be determined from evaluations of software support risk.

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SITE

SURVEY

FORM

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**Site Survey Information**

Survey data is needed to determine a historical basis of software maintenance activity across several sites which support Air Force systems.

Briefly, the survey data to be collected includes:

- a) Background data on each software system;
- b) A high-level, subjective assessment by site personnel of the adequacy (product, environment, life cycle process) of the support for each major software system;
- c) Actual software maintenance data (corrections, enhancements, conversions) for each software system in as much detail as is available. Information on each software problem corrected in each block release since "delivery" is needed. This information will be collected during the site visit.

This data will be used to determine:

- a) the availability and consistency of such data;
- b) the effort required to collect such data;
- c) the utility of the data for use in a proposed software supportability risk assessment methodology;
- d) the potential for derivation of a general data collection format for software maintenance data based upon the availability, consistency, effort, and utility as above.

Software systems for which data is desired are indicated in an accompanying list. Other suggested systems for which data might be available will be added to the list as time to collect such data permits. Typically, it should require no more than 30 minutes of a senior software person's time to complete the information for each software system. During the on-site visit it would be beneficial to talk with each of the senior personnel completing the survey form, as well as the appropriate personnel maintaining the configuration management status accounting information. In this way, problems with the assessment data can be resolved and maintenance data which is available on each system can be efficiently collected.

## DEFINITIONS FOR SITE SURVEY

ATE - Automatic Test Equipment

CSCI - Computer Software Configuration Item

IOC - Initial Operational Capability

MA - Maintenance Action

OFP - Operational Flight Program

PMRT - Program Management Responsibility Transfer

S/W - Software

**Software System** - A set of software (specifications, programs, and data) which constitutes a well-defined major function or group of functions. Typical systems include avionics OFP, ground based communications, missile guidance, simulation, threat generator, ATE, and electronic warfare.

**Software Delivery** - That point in the software life cycle when the software support function assumes responsibility for the "next" set of configuration changes to the software (e.g., next block release). This point is logically no later than PMRT, but could be as early as IOC. This applies when a contractor or government agency assumes the software support function.

**Software Life Cycle Process Management** - The policy, methodology, procedures, and guidelines applied in a software environment to the software development and support life cycle activities.

**Software Configuration Management** - A discipline applying technical and administrative direction and surveillance to (1) identify and document the functional and physical characteristics of a configuration item, (2) control changes to those characteristics, and (3) record and report change processing and implementation status.

**Software Maintenance Project Management** - The software life cycle process management applied during the support phase for the software to accomplish specific software maintenance tasks which derive from software problem reports or change requests.

**Software Maintainability** - The ease with which software can be changed in order to: correct errors, add or modify system capabilities through software changes, delete features from programs, and modify software to be compatible with hardware changes.

**Software Supportability** - A measure of the adequacy of personnel, resources and procedures to facilitate: modifying and installing software, establishing an operational software baseline, meeting user requirements.

**Support Personnel** - A general term for personnel (military, DoD civilian, or DoD contractor) whose skills are necessary to directly support mission critical system software maintenance. Includes but is not limited to management, technical, non-technical support, and contractor personnel.

**Support System** - The automated system used to change, test, or manage the configuration of mission critical system software and associated documentation. Includes but is not limited to Host Processor, Software Bench, Laboratory-Integrated Test Facility, Operation-Integrated Test Facility, and Configuration Management System.

**Support Facility** - The physical facility resources that must be available for the software support resources to accomplish a specific task(s).

**Documentation** - All of the written work describing operating and maintenance procedures for a system.

**Source Code** - The form of the program code in its source language.

**Consistency** - A measure of the extent the software products correlate and contain uniform notation, terminology, and symbology.

**Descriptiveness** - A measure of the extent that software products contain information regarding its objectives, assumptions, inputs, processing, outputs, components, revision status, etc.

**Expendability** - A measure of the extent that a physical change to information, computational functions, data storage, or execution time can be easily accomplished once the nature of what is to be changed is understood.

**Instrumentation** - A measure of the extent that software products contain aids which enhance testing.

**Modularity** - A measure of the extent that a logical partitioning of software products into parts, components, and/or modules has occurred.

**Simplicity** - A measure of the extent that software products reflect the use of singularity concepts and fundamental structures in organization, language, and implementation techniques.

**Time to Complete MA** - The time from formal notification (e.g., receipt of anomaly report or software change request) of a software maintenance request to the final disposition of that request (e.g., change is integrated into the next release, or request is denied).

**Baseline Software Supportability Profile** - The set of 27 pairs of numbers (or any subset) determined by specifying the (time to complete request, number of requests per unit time) pair for each request category. A request category is the triple (type, priority, complexity) where type is conversion, enhancement, or correction; priority is emergency, urgent, or normal; and complexity is high, medium, low.

**Emergency MA** - an MA requiring all available personnel's dedicated effort to correct the problem as soon as possible (e.g., 24 hours); MIL-STD-1679 severity code 1 or 2: mission termination or severe degradation

**Urgent MA** - an MA requiring next "block release" turnaround; MIL-STD-1679 severity code 3: mission impact

**Normal MA** - an MA not in the Emergency or Urgent categories; MIL-STD-1679 severity code 4 or 5: mission inconvenience

**High Complexity MA** - an MA where changes are in requirements, design, code, and test; or > 10% of CSCI is affected; or several modules are affected by the change (global changes); or the technical nature of the change requires highly specialized personnel skills; or the level of effort by personnel is large

**Medium Complexity MA** - an MA where changes are in design, code and test; or > 1% of CSCI is affected; or at least two modules are affected by the change (semi-local); or the level of effort by personnel is average

**Low Complexity MA** - an MA where changes are isolated to only one unit (e.g., one module/compilation unit) of code; or no more than 1% of CSCI is affected; or the level of effort by personnel is minimal

**Conversion (Adaptive) MA** - Any change/effort to a software system which is initiated as a result of changes in the environment (e.g., hardware, system software) in which the software system must operate.

**Enhancement (Perfective) MA** - Any change, insertion, deletion, modification, extension, and enhancement made to a software system to meet the evolving needs of the user.

**Corrective MA** - Any change which is necessitated by actual faults (induced or residual) in a software system.

**Risk** - The potential for realization of unwanted, negative consequences of an event.

**Software Supportability Risk** - The probability at a given point during the software support phase that the software maintenance activity specified by a baseline software supportability profile can not be accomplished with the available software support resources.

Date: \_\_\_\_\_ Site: \_\_\_\_\_ Source of Data: \_\_\_\_\_

Page 1

1. S/W BACKGROUND DATA  
(Complete for each S/W System)

1.1 IDENTIFICATION:

1.1.1 System: \_\_\_\_\_

1.1.2 S/W System: \_\_\_\_\_

1.1.3 S/W System Type (OFP,C3I,EW,Simulator,Missile,ATE): \_\_\_\_\_

1.2 DESCRIPTION:

1.2.1 Size (#CSCIs, #Modules, #Source Lines): \_\_\_\_\_

1.2.2 List Documentation Delivered by Contractor and/or  
Developed During Maintenance:  
-----  
-----

1.2.3 Language(s) & %Use: \_\_\_\_\_

1.2.4 Development Contractor Data:  
Name(s): \_\_\_\_\_

Development period: \_\_\_\_\_

Personnel Time : \_\_\_\_\_

1.2.5 Description of any Major Life Cycle Events (contractor change, major  
modification, etc.):  
-----  
-----  
-----

1.2.6 Personnel Currently Supporting S/W System:

Total number: \_\_\_\_\_

List the number by skill level (1 to 5) with 1 = Low, and 5 = High:  
#Lev1 = \_\_\_\_\_ ; #Lev2 = \_\_\_\_\_ ; #Lev3 = \_\_\_\_\_ ; #Lev4 = \_\_\_\_\_ ; #Lev5 = \_\_\_\_\_

Indicate approximate % of the time these personnel are dedicated to  
support of this S/W system: \_\_\_\_\_

1.2.7 Computer Systems Currently Supporting S/W Maintenance:  
List computers/peripherals/. or a document containing information:  
-----  
-----

Indicate approximate % of the time these systems are dedicated to  
support of this S/W system: \_\_\_\_\_

1.2.8 Software Supportability Problems:

List any significant problems which affect this system's software  
supportability.

## 2. S/W ASSESSMENT DATA

## .1 S/W PRODUCT MAINTAINABILITY ASSESSMENT

On a scale of -50 to 50 rate the S/W System Product attributes:  
 worst -> ! inadequate ! adequate ! <- best

v-----v-----v  
 -50            0            50

Do\_not\_enter\_0.                      At Delivery              Current

2.1.1 S/W Documentation : \_\_\_\_\_

2.1.1.1 Modularity : \_\_\_\_\_  
 2.1.1.2 Descriptiveness: \_\_\_\_\_  
 2.1.1.3 Consistency : \_\_\_\_\_  
 2.1.1.4 Simplicity : \_\_\_\_\_  
 2.1.1.5 Expandability : \_\_\_\_\_  
 2.1.1.6 Instrumentation: \_\_\_\_\_

2.1.2 S/W Source Code : \_\_\_\_\_

2.1.2.1 Modularity : \_\_\_\_\_  
 2.1.2.2 Descriptiveness: \_\_\_\_\_  
 2.1.2.3 Consistency : \_\_\_\_\_  
 2.1.2.4 Simplicity : \_\_\_\_\_  
 2.1.2.5 Expandability : \_\_\_\_\_  
 2.1.2.6 Instrumentation: \_\_\_\_\_

2.1.3 General S/W Maintainability : \_\_\_\_\_

## .2 S/W SUPPORT ENVIRONMENT ASSESSMENT

On a scale of -50 to 50 rate the S/W Support Environment Attributes:  
 worst -> ! inadequate ! adequate ! <- best

v-----v-----v  
 A            -50            0            50

Do\_not\_enter\_0.                      At Delivery              Current

2.2.1 S/W Support Personnel : \_\_\_\_\_

2.2.1.1 Management : \_\_\_\_\_  
 2.2.1.2 Technical : \_\_\_\_\_  
 2.2.1.3 Support : \_\_\_\_\_  
 2.2.1.4 Contractor : \_\_\_\_\_

2.2.2 S/W Support Systems : \_\_\_\_\_

2.2.2.1 Host Computer : \_\_\_\_\_  
 2.2.2.2 Software Bench : \_\_\_\_\_  
 2.2.2.3 Lab-Integ. Test: \_\_\_\_\_  
 2.2.2.4 Operational Sys: \_\_\_\_\_  
 2.2.2.5 Other (Specify): \_\_\_\_\_

2.2.3 S/W Support Facility : \_\_\_\_\_

2.2.3.1 Office Space : \_\_\_\_\_  
 2.2.3.2 System Environ.: \_\_\_\_\_

2.2.4 General S/W Support Environment: \_\_\_\_\_

**2.3 S/W LIFE CYCLE SUPPORT MANAGEMENT ASSESSMENT**

On a scale of -50 to 50 rate the S/W Life Cycle Support Management:  
worst -> ! inadequate ! adequate ! <- best  
v-----v-----v  
-50 0 50

Do not enter 0.

At Delivery Current

2.3.1 S/W Configuration Mgmt	:	-----	-----
2.3.1.1 Identification	:	-----	-----
2.3.1.2 Status Account.	:	-----	-----
2.3.1.3 Config. Control	:	-----	-----
2.3.1.4 Audit	:	-----	-----

2.3.2 S/W Maintenance Mgmt	:	-----	-----
2.3.2.1 Planning	:	-----	-----
2.3.2.2 Organization	:	-----	-----
2.3.2.3 Design Methods	:	-----	-----
2.3.2.4 Coding Methods	:	-----	-----
2.3.2.5 Test Methods	:	-----	-----
2.3.2.6 Org. Interface	:	-----	-----

2.3.3 General SWLC Support Management:	-----	-----
--	-------	-------

**2.4 S/W SUPPORTABILITY ASSESSMENT**

On a scale of -50 to 50 rate this system's overall software supportability:  
worst -> ! inadequate ! adequate ! <- best  
v-----v-----v  
-50 0 50

Do not enter 0.

At Delivery Current

2.4.1 General S/W Supportability	:	-----	-----
----------------------------------	---	-------	-------

**2.5 S/W SUPPORTABILITY RISK ASSESSMENT**

On a scale of 0(none) to 1(certain), estimate the S/W Supportability Risk for this system: that is, estimate the probability that the baseline profile of maintenance requests for this S/W System can not be completed in a unit of time (e.g., year or block release as is appropriate) given the adequacy of the software product quality, software support environment, and the software life cycle management.

2.5.1 S/W Supportability Risk	:	-----	-----
-------------------------------	---	-------	-------

### 3. DESIRABLE MAINTENANCE DATA FOR EACH SOFTWARE SYSTEM

- .1 FOR\_EACH\_BLOCK\_RELEASE\_SINCE\_SOFTWARE\_DELIVERY
  - a. List of specific software changes implemented
  - b. Estimated person (configuration management, maintenance project) effort
  - c. Actual person effort
  - d. Engineering Start and End dates
  - e. Time from Engineering End date till release was fielded
- .2 FOR\_EACH\_SOFTWARE\_CHANGE\_REQUEST\_SINCE\_SOFTWARE\_DELIVERY
  - a. Id and description
  - b. Type (correction, enhancement, conversion)
  - c. Priority (emergency, urgent, normal)
  - d. Complexity (high, medium, low)
  - e. Estimated person (configuration management, maintenance project) effort
  - f. Actual person effort
  - g. Configuration management open and close dates
  - h. Release in which change is or will be implemented
- .3 FOR\_EACH\_YEAR\_SINCE\_SOFTWARE\_DELIVERY
  - a. Number of software change requests carried over from previous year
  - b. Number of software change requests opened during current year
  - c. Number of software change requests closed during current year
- .4 ADDITIONAL\_DATA\_OF\_INTEREST
  - a. Computer system resources (e.g., computer hours) used for each release
  - b. Specific tradeoff factors which were required for each release such as request priority, personnel availability and experience, computer systems availability and adequacy
  - c. Major problems which led to delay or inefficiency in completion of a release

## **D. System Data**

## APPENDIX D

## SYSTEM DATA

## D.1 INTRODUCTION.

a. This appendix contains summaries of the raw maintenance support data gathered from the various sites visited. The sites visited include:

- (1) NORAD Space Command, Colorado Springs, CO
- (2) Warner Robins ALC, Robins AFB, GA
- (3) Sacramento ALC, Sacramento, CA
- (4) Castle AFB, CA
- (5) Ogden ALC, Ogden, UT
- (6) Oklahoma City ALC, Oklahoma City, OK
- (7) Langley AFB, VA.

At each site, maintenance support data for several systems were collected. Each system (e.g., F-16 at Ogden ALC) generally had several software systems (e.g., FCC, SMS, RDR, HUD). For each software system the maintenance support data consisted of background data, evaluation data, and maintenance activity data on each block release since the beginning of formal software system support activity at the site.

b. The terminology developed to describe the data in a consistent way across software systems is described in section D.2 of this

appendix and appendix B, Glossary of Terms. The actual data collected, and in some cases interpreted from notes and application of the terminology constraints, are summarized in section D.3 of this appendix. The bulk of the analysis results presented in this report is derived from the data presented in this appendix.

## D.2 TERMINOLOGY.

The Glossary of Terms, appendix B, contains reasonably concise definitions for the terms used in this report. However, there are some caveats relative to the manner in which the actual data is "molded" into the appropriately defined terms. This section is a brief attempt to describe those caveats for the specific data items used in section D.3 of this appendix.

### D.2.1 Background Data.

a. There were considerable background data collected during the individual interview sessions and from the data survey forms. The more important background data (across software systems) are summarized by software system in section D.3. The Program Management Responsibility Transfer (PMRT) date is officially when organic software system support is supposed to begin. Many systems have not undergone PMRT, but some have already begun the software support function. In this case, the "delivery" date reflects this unofficial beginning of the support function.

b. In some cases the organic support is a combination of several organic organizations and/or perhaps a contractor. As much as possible, the personnel counts reflect actual maintenance support personnel, not the personnel which may be part of an ALC overhead management function or a contractor function required in order to process an "official" release, because the software system has not officially undergone PMRT. These actual software maintenance

personnel are the management, technical, support, and contractor personnel directly involved in the configuration management and/or analysis, design, code, test of changes in a block release.

c. Major problem areas reflect the particular opinions of the personnel interviewed, and may not reflect the opinion of other management on-site personnel.

#### D.2.2 Evaluation Data.

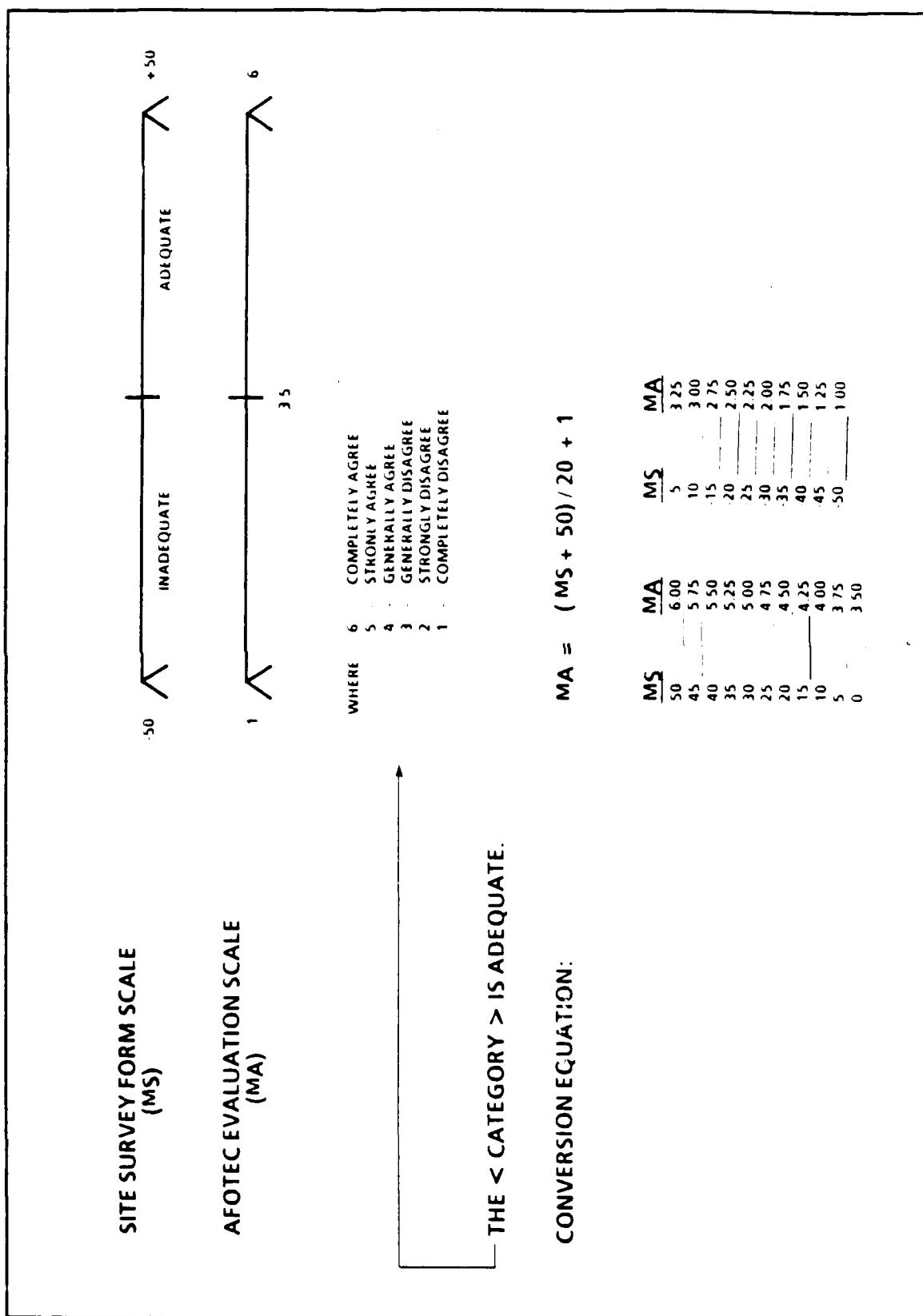
a. The evaluation data for each software system represented the subjective opinion of the personnel completing each data survey form. Each value represented the adequacy of the evaluated supportability category on a scale from -50 (totally inadequate) to +50 (totally adequate). The "0" value is the separator of "inadequate" and "adequate". The transformation of values to AFOTEC's evaluation scale 1 to 6 is illustrated in table D-1.

b. As an example of how to use table D-1, suppose the evaluated score of the category S/W source code modularity is a 20. Then the corresponding AFOTEC score would be 4.5. This score (4.5) would correspond approximately to a value midway between "generally agree" and "strongly agree" values for the statement: "The modularity of the source code is adequate."

c. The primary concern of the evaluation was to determine how the supportability metrics compare (correlate) with the concept of supportability risk. Unfortunately, the explanation of risk seemed to be misinterpreted by many evaluation personnel. This has led to a more precise statement of supportability risk as defined in the glossary of terms. As applied to the site survey baselines, the supportability risk is "the probability that the specified block release cannot be accomplished within the available software support resources." "At delivery" the block release would be the first block

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Table D-1. Survey Form Evaluation Score Conversion.



release. For "current" the block release would be the current one being processed or, if none were being processed, the next expected block release.

d. The focus of supportability risk is upon the risk to complete the agreed upon changes in a block release as opposed to all submitted change requests. This focus is required because very little data exist concerning submitted change requests, except for the consensus that there will "always" be more change requests than could possibly be processed. The backlog estimates range from 20 to 200 percent at the current release change count.

e. Thus, the supportability risk being estimated is the risk of being unable to complete a block release once the contents of a block release have been essentially agreed upon during preliminary analysis. "Unable to complete" is still a fuzzy term, but it includes such things as changes being added and/or additional resources (such as personnel, calendar time, support tools) being required. If the user or any other personnel changes the scope of the block release content in such a manner that the block release will be late or more resources must be added to keep the schedule, then the original block release was not completed as agreed upon. The possibility of this happening is the supportability risk.

f. It is clearly realized that there is more to supportability risk than is being measured by this data. However, it does appear that the concept of baseline maintenance support activity (changes in a block release), supportability factors (software products, software support environment, software life cycle management), supportability factor metrics, and supportability risk (as defined here) are reasonably consistent and related terms.

D.2.3 Maintenance Activity Data.

a. Maintenance activity data are the set of all productivity information concerning each block release of changes to a software system. The data upon which this report focus include:

- (1) Release start and engineering completion dates
- (2) Number of personnel available for direct support of the block release
- (3) Percentage of time these personnel are dedicated to this software system
- (4) Personnel overlap factor with other releases
- (5) Number of changes in release
- (6) Number of changes by type (correction, enhancement, conversion)
- (7) Number of changes by complexity (low, medium, high)
- (8) Number of changes by priority (normal, urgent, emergency).

b. The release start date is that date when analysis activity related to the subject block release begins for which support personnel are required. Typically, this might be the date of the first change request or perhaps the date when no more change requests are accepted for consideration. The engineering completion date is that date when the engineering (including operational testing) part of the block release is complete. Time for "kit" proofing, prom burning, and creation of technical orders after completion of engineering is

not included. There is usually additional time between the engineering completion date, and the actual fielded date. In fact, an engineering release may never be fielded.

c. The number of personnel is the count of those persons assigned in some direct capacity to the support of the software system. It could be management, technical, support (technicians, librarian clerks), or contractor personnel. The percentage of time these people are dedicated to this software system as opposed to other software systems is required in order to determine "full time equivalent" personnel available to support the software system. This percentage dedicated does include time spent by these personnel performing various "overhead" functions even if not directly related to the software system. Thus, full time equivalent personnel time does include certain overhead time not directly devoted to software maintenance activity. Such time would include:

- (1) Vacations and sick leave
- (2) Supporting outside interests such as test agencies and user meetings
- (3) Support of internal site functions such as internal meetings, and organization training.

d. In addition to the available "full time equivalent" personnel, it is necessary to account for any overlap by the same personnel in supporting consecutive releases. If consecutive releases involve no overlap, then this factor is 1.0. If 50 percent of the time is spent on each of the releases, then an overlap factor of 0.5 is used for each release. In general, if the release dates (start and end) overlap for consecutive releases, then an overlap factor of 0.5 for the duration of the overlapped time has been used. In other cases, time may have been spent against a planned release which is not completed

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and thus never shows up as an overlap. This is totally subjective and can only be accurately specified by the personnel familiar with the given release. Updates to the specified overlap factor (as well as any other data) will be solicited through a normal delphi technique with the software personnel who were the primary source for this data.

e. The total number of changes in each release is very accurate (except in a few rather obvious instances), and does represent the number of official, documented, change requests (MIP, SPRs, DRs, SMRs, and so forth, as appropriately named by the system's configuration control procedures). The change request generally initiated individual analysis, design, code, and test as well as integrated block release analysis and test. The resulting changes to the software system might be to one module or might be to many modules. The changes to documentation and source code might involve everything from changes to requirements, to simple one-line parameter updates. Just because a change involves only one module does not necessarily imply it is simple. The nature of the change (e.g., development of a state-of-the-art EW algorithm) might dictate much analysis and design, but little code change. The complexity of the change is subjectively defined in terms of scale values high, medium, low in accordance with the combination of skill level of resources required, amount of software product affected, and amount of resources (personnel and support system) required by the change request.

f. Generally, the number of conversions was not delineated from number of enhancements in the data. Although it was clear from the interviews that much conversion activity is being done, the conversions are usually included with enhancements and are not easily separable.

g. Except for NORAD, the other sites (primarily ALCs) had only NORMAL (i.e., routine) priority assigned to the change requests.

Concern among ALC personnel for possible problems in adequate response to non-normal priority change requests was indicated. The issue of processing security sensitive changes which fall outside of the "normal" request priority was also raised several times. An estimate of 2 to 2 1/2 times normal change processing time was given for sensitive changes.

### D.3 SURVEY DATA BY SITE.

a. The raw survey data are summarized in this section by site. For each site, the data for each software system consist of background data, evaluation data, and maintenance activity data.

b. Table D-2 contains a list of the sites and software systems for which data are included along with the application type of the software system. This table has entries for 81 separate software systems.

c. Table D-3 contains the software systems background raw data and corresponds roughly to the information requested in section 1.2 of the site survey form (appendix C). These data are reported in six parts. Part 1 is a summary of the data on size in terms of Computer Software Configuration Items (CSCIs), modules, and number of source lines in thousands (k). Part 2 is a list of the primary, secondary, tertiary, and other programming languages in which the software system is written. Approximate percent of source is listed for each language. The dominant language is clearly assembler. Part 3 is a list of system development data in the form of development contractor, development period, and person years of effort. Most of the calendar and effort data in part 3 are approximate. Part 4 is a

summary of the number of personnel assigned to the system, an approximate skill rating from 1 (low) to 5 (high), and an approximate percentage of the time the assigned personnel are dedicated to the subject software system as opposed to another software system. The skill level generally reflected a level of experience with the subject software. Part 5 is a partial list of the support systems for the software and the percentage of time the support systems are dedicated to the subject software. Part 6 lists software supportability problems reported by the support personnel interviewed during the survey visit.

d. Table D-4 contains the software supportability evaluation data and corresponds to the information requested in section 2 of the site survey form (appendix C). This table is separated into six parts. The first three parts correspond to the software product, software support facility and software support life cycle management evaluation data for the "AT DELIVERY" system. The latter three parts correspond to the similar evaluations for the "CURRENT" system. Raw data values of -99 indicate data are missing. Raw data values of 99 indicate the category was not applicable for the subject system. All categories in the software product and life cycle management evaluations are applicable. Only a few in the software support facility evaluation (e.g., contractor personnel, "other" support system, and perhaps one of the support system environments) are possibly not applicable. There may be a few typographical errors in the data as to use of the 99 and -99 values, but all other data have been validated against the information entered on the site survey form. Note that there is generally one evaluation per software system. For the F-4 software systems, multiple evaluations were done. This will be helpful for future analysis efforts.

e. Table D-5 contains the software maintenance activity data as reduced for commonality across software systems. Some of these data are most subjective and need to be reviewed carefully by the cognizant support personnel to improve accuracy.

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f. In particular, the various counts (total, type, complexity, priority) are reasonably accurate if they exist. A zero for all fields of type, complexity or priority indicates missing data. These data would be very helpful if they could be obtained.

g. The release start date and engineering completion date correspond to the release duration in months. Many of these dates are best guesses. An improvement in accuracy would be a major improvement in computation of the profile charts.

h. The number of personnel and the percentage of time dedicated to the software system are essentially directly from the background data (section 1.2.6 in the site survey form). Occasionally, these data were missing or conflicted with information obtained during an interview. In these cases a best guess was attempted. An improvement in accuracy of these data would be a major improvement in computation of the profile charts.

i. The percent dedicated to the release is doubly subjective because it depends upon the accuracy of the release overlaps, and the assumption that, given an overlap, the sharing of personnel is distributed evenly (for each release across an overlap time period). In addition, some of the "quicky" interim/urgent/emergency releases were difficult to categorize. The intent of this percent factor was to reduce, in a reasonably logical and consistent manner, the person time allocated against a given release when the same personnel were being used across several releases for the software. As an example, the NORAD software system releases were overlapped at least three to a year over approximately 11-month release cycles. Any better estimate for this factor would also be a major improvement in computation of the profile charts.

j. The data in tables D-3, D-4, and D-5 represent a wide variety of interesting information. The current analysis is based upon these

data. The future analysis to be included in the final draft report will be based upon these data along with as much improved data as can be solicited from the original evaluators and support personnel interviewed.

Table D-2. Sites and Software Systems

ID	SITE	SYSTEM	SOFTWARE SYSTEM	SOFTWARE TYPE
1	NORAD	CSS	CSE	
2	NORAD	MDS	CSE	
3	NORAD	MEBU	CSE	
4	NORAD	NCS	CSE	
5	NORAD	SSC	SSC	
6	WR-ALC	ALR-46	ALR-46	
7	WR-ALC	ALR-69	ALR-69	
8	WR-ALC	AN/ALQ-131	AGEOP	
9	WR-ALC	AN/ALQ-131	BIG	
10	WR-ALC	AN/ALQ-131	OFF	
11	WR-ALC	AN/ALQ-131	UUJ	
12	WR-ALC	AN/ALQ-131	UUE	
13	WR-ALC	AFR-38	APR-38	
14	WR-ALC	B-52 EVS ATE	ASD-151	AIE
15	WR-ALC	E-3A AVIONICS ATE	AN/GSM-285(B)	AIE
16	WR ALC	E-3A AVIONICS ATE	AN/GSM-285(W)	AIE
17	WR-ALC	F-15	CC	OFF
18	WR-ALC	F-15 AVIONICS ATE	RADAR	OFF
19	WR-ALC	WR-ALC	ADIS,AIS	AIE
20	WR-ALC	JTIDS	ASIT/DCLP	LIE
21	WR-ALC	JTIDS	E-3A AWACS/DCP	C
22	WR-ALC	JTIDS	SP/USER	9UM
23	WR-ALC	JTIDS	SYS EXERCISER	SIM
24	WR-ALC	PAVE TACK	AISF	SIM
25	SM-ALC	PAVE TACK	OFF	OFF
26	SM-ALC	F-111D	WNC	OFF
27	SM-ALC	F-111F	WNC	OFF
28	CASILE AFB	FB-111A	WNC	OFF
29	CASILE AFB	B-52	EPT	AID
30	CASILE AFB	KC-135	WST	AID
31	CASILE AFB	T-4 TRAINER	WSI	AID
32	DO-ALC	F-16	T-4 SIMULATOR	AID
33	DO ALC	F-16	FCC	OFF
34	DO ALC	F-16	HUD	OFF
35	DO ALC	F-16	OFF	AID
36	DO ALC	F-16	FCR	OFF
37	DO ALC	F-16	SMS	OFF
38	DO ALC	F-4	MD16	OFF
39	DO ALC	F-4E	AN/ARN-101	OFF
40	DO ALC	F-4G	AN/ARN-101	OFF
41	DO ALC	F-4G	IRU-1/ACH	OFF
42	DO ALC	MINUTEMAN	WING 11/2015	SIM
43	DO ALC	MINUTEMAN	WING V1/VIS 29	SIM
44	DO ALC	MINUTEMAN II	WINGS/VIS 28	SIM
45	DO ALC	MINUTEMAN II	SSAS/CAVS	SIM
46	DO ALC	MINUTEMAN II	WING V1/VIS/RAIS	SIM
47	DO ALC	RD-4C	WING V1/VIS/RAIS	SIM
48	DO ALC	RD-4C	AN/ARN-101	OFF
49	DO ALC	RD-4C	U1	AID
50	DO ALC	RD-4C	U1	OFF
51	DO ALC	RD-4C	U1	OFF

Table D-2. Sites and Software Systems

ID	SITE	SYSTEM	SOFTWARE SYSTEM	SOURCE TYPE
52	OC-ALC	B-1B	C119	OFF
53	OC-ALC	B-1B	EMUX	OFF
54	OC-ALC	B-1B	F/CMS	OFF
55	OC-ALC	B-1B	INS	OFF
56	OC-ALC	B-1B	ORS	OFF
57	OC-ALC	B-52	BNSI	AID
58	OC-ALC	B-52	F1SS	SUP
59	OC-ALC	B-52	MC-1 EXEC	OFF
60	OC-ALC	B-52	MC-2 EXEC	OFF
61	OC-ALC	E-3A	INS	OFF
62	OC-ALC	E-3A	OMEGA	OFF
63	OC-ALC	E-3A	SMCF	OFF
64	OC-ALC	E-3A	SRCP	OFF
65	OC-ALC	E-3A	SIGSLP	SUP
66	OC-ALC	GLCM	DFS	SUP
67	OC-ALC	GLCM	H DID	SUP
68	OC-ALC	GLCM	MFT	SUP
69	OC-ALC	GLCM	OFF	OFF
70	OC-ALC	GLCM	WCS	OFF
71	OC-ALC	GRAM	OFF	OFF
72	TINKER AFB	E-3A	AUCF	C-E
73	TINKER AFB	E-3A	UTILITIES	SUP
74	LANGLEY	J11DS	ASII/IPOCP	C-E
75	LANGLEY	SIRIS	STRIS	AID
76	LANGLEY	TACS	CAIMS	C-E
77	LANGLEY	TIFI	DC/SR	C-E
78	LANGLEY	TIP1	LI/MARKES/TEREC	C-E
79	LANGLEY	4071	HUGHES UTIL	SUP
80	LANGLEY	4071	IBM UTIL	SUP
81	LANGLEY	407L	IOPF/IMPP	C-E

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Table D-5. Systems Background Raw Data  
Part I: SYSTEM SIZE

ID	SITE	SYSTEM	SOFTWARE SYSTEM	EVALUATION DATE	#CSC's	#MODULES	#SOURCE LINES (K)
1	NORAD	SSB	CSS	01/09/85	16	0	350
2	NORAD	MEBU	MEBU	01/09/85	3	0	123
3	NORAD	NCS	NCS	01/09/85	4	446	231
4	NORAD	SSC	SSC	01/16/85	20	3600	1000
5	WKA-LC	ALR-46	ALR-46	01/29/85	1	17	17
6	WKA-LC	ALK-69	ALK-69	01/30/85	2	42	32
7	WKA-LC	AN/ALD-151	BTG	01/29/85	0	0	300
8	WKA-LC	AN/ALD-131	OFF	01/29/85	0	0	112
9	WKA-LC	AN/ALD-151	UNIT	01/29/85	0	0	400
10	WKA-LC	APR-38	APR-38	01/29/85	2	20	120
11	WKA-LC	B-52 EVS AIE	ASU-151	01/29/85	15	160	250
12	WKA-LC	E-5A AVIONICS ATE	AN/GSM-285 (B)	01/31/85	70	95	200
13	WKA-LC	E-3A AVIONICS ATE	AN/GSM-285 (W)	01/31/85	340	500	1000
14	WKA-LC	F-15	LC	01/31/85	1	67	0
15	WKA-LC	F-15	RADAR	01/31/85	0	0	39
16	WKA-LC	F-15 AVIONICS ATE	AUTS, AIS	01/31/85	0	256	2600
17	WKA-LC	JTIDS	ASIT/OCP	01/31/85	1	231	37
18	WKA-LC	JTIDS	E-3A AWACS/OCP	01/31/85	1	237	37
19	WKA-LC	JTIDS	SP/USER	01/31/85	1	166	26
20	WKA-LC	JTIDS	SYS EXERCISER	01/31/85	3	633	225
21	WKA-LC	FAVE TACI	ALSF	01/31/85	3	100	75
22	WKA-LC	FAVE TACI	UFP	02/01/85	5	30	13
23	SM-ALC	F-111D	WNC	02/26/85	2	26	40
24	SM-ALC	F-111F	WNC	03/01/85	2	30	40
25	SM-ALC	FB-111A	WNC	02/28/85	2	30	36
26	CASTLE AFB	B-52	CPI	02/21/85	0	163	100
27	CASTLE AFB	B-52	WSI	02/21/85	0	1000	1000
28	CASTLE AFB	F-1C-1, 25	WST	02/21/85	0	2000	500
29	CASTLE AFB	F-4 TRAINER	1-4 SIMULATOR	02/21/85	0	140	20
30	DO-ALC	F-16	FCC	04/24/85	12	150	32
31	DO-ALC	F-16	HUD	04/24/85	0	127	16
32	DO-ALC	F-16	UFT	04/24/85	0	1000	10
33	DO-ALC	F-16	FUR	04/24/85	0	0	64
34	DO-ALC	F-16	SMS	04/24/85	0	143	50
35	DO-ALC	F-4	MOTS	04/24/85	0	0	59
36	DO-ALC	F-4	MOTS	04/24/85	0	0	60
37	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	50	50
38	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
39	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
40	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
41	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
42	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
43	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
44	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
45	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
46	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
47	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
48	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
49	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50
50	DO-ALC	F-4E	AN/ARN-101	04/24/85	0	0	50

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104-105Table D-3. Systems Background Raw Data  
Part I: SYSTEM SIZE

ID	SITE	SYSTEM	SOFTWARE SYSTEM	EVALUATION DATE	CCS/C	MODULES (STARTING LINE #)
51	OO-ALC	MINUTEMAN	WING 11/2015	04/24/85	50	40
52	OO-ALC	MINUTEMAN	WING VI/HB-29	04/24/85	75	75
53	OO-ALC	MINUTEMAN	WINGS/HB-28	04/24/85	60	60
54	OO-ALC	MINUTEMAN II	SSAS/LAPS	04/29/85	10	10
55	OO-ALC	MINUTEMAN II	WING V/HBG/RATS	04/29/85	30	30
56	OO-ALC	MINUTEMAN II	WING VI/HBG/RATS	04/29/85	30	30
57	OO-ALC	RF-4C	AN/ARN-101	04/24/85	0	49
58	OO-ALC	RF-4C	AN/ARN-101	04/25/85	0	44
59	OO-ALC	RF-4C	AN/ARN-101	04/24/85	0	49
60	OO-ALC	RF-4C	AN/ARN-101	04/24/85	0	49
61	OO-ALC	RF-4C	AN/ARN-101	04/24/85	0	49
62	OO-ALC	RF-4C	AN/ARN-101	04/24/85	0	49
63	UC-ALC	AL LN	LIT	05/15/85	32	BB
64	UC-ALC	AL CR	LIT	05/14/85	4	15
65	OC-ALC	AL LN	OFP	05/14/85	1	41
66	OC-ALC	AL LN	OFP	05/16/85	1	41
67	UC-ALC	B-1B	CADC	05/15/85	1	12
68	UC-ALC	B-1B	LITS	05/14/85	42	162
69	UC-ALC	B-1B	EMUX	05/15/85	1	18
70	DC-ALC	B-1B	F/LGMS	05/13/85	1	15
71	DC-ALC	B-1B	INS	05/14/85	204	30
72	DC-ALC	B-1B	DRS	05/13/85	1	18
73	DC-ALC	B-52	BNSI	05/15/85	0	1
74	DC-ALC	B-52	FTSS	05/20/85	0	45
75	DC-ALC	B-52	MC-1 EXEC	05/14/85	0	70
76	DC-ALC	B-52	MC-2 EXEC	05/15/85	0	70
77	DC-ALC	E-34	INS	05/15/85	0	14
78	DC-ALC	E-34	OMEGA	05/14/85	0	16
79	DC-ALC	E-34	SMCF	05/14/85	5	68
80	DC-ALC	E-34	SKCF	05/16/85	0	450
81	DC-ALC	E-34	SKGSCH	05/14/85	4	31
82	DC-ALC	AL CR	DPS	05/15/85	0	73
83	DC-ALC	AL LN	M-D10	05/15/85	0	99
84	UL-HL	AL LN	MF-1	05/15/85	0	91
85	UL-HL	AL LN	OFP	05/15/85	0	32
86	UL-HL	AL LN	WCS	05/15/85	0	126
87	DC-ALC	SKAF	OFP	05/16/85	0	18
88	LINKER AFb	E-34	AOCF	05/14/85	273	273
89	LINKER AFb	E-34	UTILITIES	05/14/85	0	0
90	LINKER	J1105	ASII/FUCP	07/23/85	18	100
91	LINKER	S1H15	STATS	07/24/85	3	262
92	LINKER	TAC-2	CAFM	07/23/85	0	0
93	LINKER	TAC-1	DC/SR	07/23/85	0	0
94	LINKER	TAC-1	LL/MARSH/TEREC	07/23/85	2	280
95	LINKER	4111	HUGHES UTIL	07/24/85	0	609
96	LINKER	4111	IBM UTIL	07/24/85	0	118
97	LINKER	4111	FORF/IMF	07/24/85	0	241

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Table B-5. Systems Background Raw Data  
Part 2: PROGRAMMING LANGUAGES

ID SITE	SYSTEM	SW/WARE SYSTEM	LANGUAGE1	LANGUAGE2	LANGUAGES	OTHER LANGUAGES
1 NORAD	LSS	LSS	ASSEMBLY	JOVIAL	JOVIAL	JOVIAL
2 NORAD	MEBU	MEBU	ASSEMBLY	69 ASSEMBLY	69 ASSEMBLY	ECI, DRL
3 NORAD	NCS	NCS	FORTran	49 JOVIAL	49 JOVIAL	/ COMP/UPK1/ASSLM, FORT, JCL
4 NORAD	SSC	SSC	ASSEMBLY	BU JOVIAL	BU JOVIAL	ASSEMBL 1
5 WH-AU-L	AIR-4b	AIR-4b	ASSEMBLY	100 CUBUL	100 CUBUL	0
6 WH-AU-L	ALK-8Y	ALK-8Y	ASSEMBLY	100 CUBUL	100 CUBUL	0
7 WH-AU-L	AN/ALU-151	AN/ALU-151	ATLAS	100 CUBUL	100 CUBUL	0
8 WH-AU-L	AN/ALD-151	AN/ALD-151	ASSEMBLY	100 CUBUL	100 CUBUL	0
9 WH-AU-L	AN/NAQ-151	AN/NAQ-151	ASSEMBLY	100 CUBUL	100 CUBUL	0
10 WH-AU-L	APR-3B	APR-3B	ASSEMBLY	100 CUBUL	100 CUBUL	0
11 WH-AU-C	IS-52 EVS AIR	IS-52 EVS AIR	FORTRAN	50 ASSEMBLY	50 ASSEMBLY	0
12 WH-AU-C	E 52 AVIONICS AIR	E 52 AVIONICS AIR	FORTRAN	65 LASAR FFG	65 LASAR FFG	0
13 WH-AU-C	E 52 AVIONICS AIR	E 52 AVIONICS AIR	FORTRAN	67 LASAR FFG	67 LASAR FFG	0
14 WH-AU-C	F 15	F 15	ASSEMBLY	100 CUBUL	100 CUBUL	0
15 WH-AU-C	F 15 AVIONICS AIR	F 15 AVIONICS AIR	ASSEMBLY	100 CUBUL	100 CUBUL	0
16 WH-AU-C	F 15 AVIONICS AIR	F 15 AVIONICS AIR	FORTRAN	10 DAF	10 DAF	0
17 WH-AU-C	JILLUS	JILLUS	ATLAS	100 CUBUL	100 CUBUL	0
18 WH-AU-C	JILLUS	JILLUS	ASSEMBLY	100 CUBUL	100 CUBUL	0
19 WH-AU-C	JILLUS	JILLUS	ASSEMBLY	100 CUBUL	100 CUBUL	0
20 WH-AU-C	JILLUS	JILLUS	ASSEMBLY	100 CUBUL	100 CUBUL	0
21 WH-AU-C	JILLUS	JILLUS	ASSEMBLY	100 CUBUL	100 CUBUL	0
22 WH-AU-C	KAVE	KAVE	FORTRAN	50 DFI	50 DFI	0
23 WH-AU-C	KAVE	KAVE	FORTRAN	100 DFI	100 DFI	0
24 SW-AU-C	F 111D	F 111D	ASSEMBLY	100 DFI	100 DFI	0
25 SW-AU-C	F 111F	F 111F	ASSEMBLY	100 DFI	100 DFI	0
26 SW-AU-C	FH 111W	FH 111W	ASSEMBLY	100 DFI	100 DFI	0
27 (ASBITE eff B 21)	(ASBITE eff B 21)	(ASBITE eff B 21)	ASSEMBLY	5 SPET.	5 SPET.	0
28 LADDALE eff B 11	LADDALE eff B 11	LADDALE eff B 11	FORTRAN	75 ASSEMBLY	75 ASSEMBLY	0
29 LADDALE eff B 11	LADDALE eff B 11	LADDALE eff B 11	ASSEMBLY	70 ASSEMBLY	70 ASSEMBLY	0
30 UA-AU-C	F 15	F 15	FORTRAN	100 DFI	100 DFI	0
31 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
32 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
33 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
34 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
35 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
36 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
37 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
38 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
39 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
40 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
41 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
42 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
43 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
44 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
45 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
46 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
47 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
48 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
49 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
50 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
51 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
52 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
53 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
54 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
55 UA-AU-C	F 16	F 16	ASSEMBLY	100 DFI	100 DFI	0
56 JUVIA						
57 FORTAN						
58 COBOL						
59 CUBUL						
60 SPET.						
61 SPETL						
62 DAF						
63 AFAN						
64 LASAR FFG						
65 LASAR FFG						
66 LASAR FFG						
67 LASAR FFG						
68 LASAR FFG						
69 LASAR FFG						
70 LASAR FFG						
71 LASAR FFG						
72 LASAR FFG						
73 LASAR FFG						
74 LASAR FFG						
75 LASAR FFG						
76 LASAR FFG						
77 LASAR FFG						
78 LASAR FFG						
79 LASAR FFG						
80 LASAR FFG						
81 LASAR FFG						
82 LASAR FFG						
83 LASAR FFG						
84 LASAR FFG						
85 LASAR FFG						
86 LASAR FFG						
87 LASAR FFG						
88 LASAR FFG						
89 LASAR FFG						
90 LASAR FFG						
91 LASAR FFG						
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Table D-3. System Background Data  
Part 2: PROGRAMMING LANGUAGES

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Table U-3. Systems Background Raw Data  
Part 3: System DEVELOPMENT

ID	SIT	SYSTH	SUB INHABE SYSTEM	DEVELOPMENT CONTRACTS	DEVELOPMENT PERIOD	EFF(kt) (P-V)
1	NURAD	L-15	CSS	FORD AEROSPACÉ NONE, DEVELOPED BY AIR FORCE	1973 - 1974	0
2	NURAD	PIR BU	PIR BU	NONE, DEVELOPED BY AIR FORCE	1973 - 1974	0
3	NURAD	NCS	NCS	NONE, DEVELOPED BY AIR FORCE	1973 - 1974	0
4	NURAD	SSC	SSC	DALMO VICTOR	1979-1-1981	5
5	WH-AFC	AIR-4B	AIR-4B	DALMO VICTOR	1979-1-1981	5
6	WH-AFC	AIR-6Y	AIR-6Y	DALMO VICTOR	1979-1-1981	5
7	WH-AFC	AN/AU-151	AN/AU-151	WESTINGHOUSE	2 YRS., FMR 10/79	0
8	WH-AFC	AN/AU-151	AN/AU-151	WESTINGHOUSE	2 YRS., FMR 10/79	0
9	WH-AFC	AN/AU-151	AN/AU-151	WESTINGHOUSE	2 YRS., FMR 10/79	0
10	WH-AFC	AN/K-3B	AN/K-3B	MAC AIR, IBM, LORCA, TI	1971-1974	0
11	WH-AFC	B-52 EVS AIE	ASU-151	BOEING	1971-1974	0
12	WH-AFC	E SA AVIATION AIE	AN/GSM-285(W)	BOEING	1978-1983	0
13	WH-AFC	E SA AVIATION AIE	AN/GSM-285(W)	USAF	NOV 78 UC1 Bb	0
14	WH-AFC	F-15	F-15	MCDONNELL DOUGLAS AIRLAR	1972-1-1975	200
15	WH-AFC	F-15	F-15	HUGHES AIRCRFT	111 YRS. (WHEN?)	1000
16	WH-AFC	F-15	F-15	MCDONNELL DOUGLAS	1972-1-1978	0
17	WH-AFC	J-11Ds	J-11Ds	HUGHES AIRCRFT	1976-1-1982	0
18	WH-AFC	J-11Ds	J-11Ds	HUGHES AIRCRFT	1976-1-1982	0
19	WH-AFC	J-11Ds	J-11Ds	HUGHES AIRCRFT	1976-1-1982	0
20	WH-AFC	J-11Ds	J-11Ds	HUGHES AIRCRFT	1976-1-1982	0
21	WH-AFC	J-11Ds	J-11Ds	HUGHES AIRCRFT	1976-1-1982	0
22	WH-AFC	F-14	F-14	FORD AEROSPACE	1969 -	0
23	WH-AFC	F-14	F-14	ROCKWELL INTL AUTOMATICS	1969	0
24	WH-AFC	F-14F	F-14F	GENERAL DYNAMICS	1969	0
25	WH-AFC	F-14	F-14	GENERAL DYNAMICS	1969	0
26	WH-AFC	F-52	F-52	SPEERY SECOK ( )	1976-1-1977	0
27	WH-AFC	F-52	F-52	SINGER, AAI	1979-1-1983 (PMK-1)	0
28	WH-AFC	F-10	F-10	SINGER	1979-1-1983 (PMK-1)	0
29	WH-AFC	F-10	F-10	HUGHES AIRCRFT	0	0
30	WH-AFC	F-10	F-10	GENERAL DYNAMICS	0	0
31	WH-AFC	F-10	F-10	MARCONI	0	0
32	WH-AFC	F-10	F-10	SINGER-LIN	0	0
33	WH-AFC	F-10	F-10	WESTINGHOUSE	0	0
34	WH-AFC	F-16	F-16	GENERAL DYNAMICS	0	0
35	WH-AFC	F-4	F-4	LEAK SIEGLER	1977-1984	0
36	WH-AFC	F-4	F-4	LEAK SIEGLER	1977-1-1984	0
37	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
38	WH-AFC	F-4E	F-4E	LEAK SIEGLER	5 YRS. (WHEN?)	0
39	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
40	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
41	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
42	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
43	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
44	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
45	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
46	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
47	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
48	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
49	WH-AFC	F-4E	F-4E	LEAK SIEGLER	1972-1-1983	0
				WESTINGHOUSE	4 yrs.	0
				WESTINGHOUSE	4 yrs.	0

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ID	Site	System	SOFI MARK SYSTEM		DEVELOPMENT CONTRACTORS		DEVELOPMENT PERIOD	EFFECTIVE PERIOD
			Part	Part	Part	Part		
51	DO A/L	E-4U	M/NUTEMAN	WING	WESTINGHOUSE	4 YRS.	0	
51	DO A/L		M/NUTEMAN	WING	5 MUS.	5 MUS.	1	
52	DO A/L		M/NUTEMAN	WING /HS-29	10 MUS.	4 MUS.	1	
53	DO A/L		M/NUTEMAN	SSB/CAPS				
54	DO A/L		M/NUTEMAN	WING V/HED/KAIS	RUL WELL	1970-1-72		
55	DO A/L		M/NUTEMAN	WING	RUL WELL	1975-1-78		
56	DO A/L		M/NUTEMAN	WING	RUL WELL	1975-1-78		
57	DO A/L		HF-4L	HF-4L	LEAK STELER	5 YRS. (WHEN?)		
58	DO A/L		HF-4L	HF-4L	LEAK STELER	1972-1-783		
59	DO A/L		HF-4C	HF-4C	LEAK STELER			
60	DO A/L		HF-4C	HF-4C	LEAK STELER			
61	DO A/L		HF-4L	HF-4L	LEAK STELER			
62	DO A/L		HF-4L	HF-4L	LEAK STELER			
63	DL A/L		DL CM	LJ I	BUT INS	1979-1-781		
64	DL A/L		DL CM	LPI	BOEING	1979-1-781		
65	DL A/L		DL CM	UF F	BOEING	1981		
66	DL A/L		DL CM	UF F	BOEING	1981		
67	DL A/L		DL CM	UADC	AIR RESEARCH	1978-1-789		
68	DL A/L		DL CM	C115	C115	1978-1-789		
69	UC -A/L		UC -A/L	UC -A/L	FUCH WELL	1978-1-789		
70	UC -A/L		UC -A/L	UC -A/L	KUCH WELL	1978-1-789		
71	UC -A/L		UC -A/L	UC -A/L	SIMMONDS	1978-1-789		
72	UC -A/L		UC -A/L	UC -A/L	SINGER	1978-1-789		
73	UC -A/L		UC -A/L	UC -A/L	WESTINGHOUSE	1978-1-789		
74	UC -A/L		UC -A/L	UC -A/L	BUT INS	2 yrs.		
75	DL A/L		DL CM	FTTS	FTTS	1983-1-85		
76	DL A/L		DL CM	MC-1 EXEC	BOEING	1990-1-782		
77	UC -A/L		UC -A/L	MC-2 EXEC	BOEING	1992-1-786		
78	UC -A/L		UC -A/L	INS	DELCO ELECTRONICS			
79	UC -A/L		UC -A/L	ONEGA	NORTHUP			
80	UC -A/L		UC -A/L	SMCP	BOEING / SEATTLE			
81	UC -A/L		UC -A/L	SKCP	LOCKHEED	1983-1-775		
82	UC -A/L		UC -A/L	DF-S	KOM, VITRO, SOC, McD- DOUGLAS	1982-1-785		
83	UC -A/L		UC -A/L	M-D	GENERAL DYNAMICS	1979-1-785		
84	DC -A/L		DC -A/L	M-T	MDONNELL - DOUGLAS	1979-1-785		
85	DC -A/L		DC -A/L	UF F	VITRO	1979-1-785		
86	DL A/L		DL A/L	WF S	BOEING	1983-1-785		
87	DL A/L		DL A/L	SHAM	BOEING	1979-1-786		
88	DL A/L		DL A/L	E-34	BOEING	1979-1-786		
89	DL A/L		DL A/L	E-34	BOEING	1979-1-786		
90	DL A/L		DL A/L	JTFD	IBM FSD (OMEGO)	1979-1-784		
91	DL A/L		DL A/L	SIRIS	GTE	1977-1-784		
92	DL A/L		DL A/L	TCPS	USAT	1981		
93	DL A/L		DL A/L	TCPS	SYSTEM DESIGN	1973-1-777		
94	DL A/L		DL A/L	TCPS	TEK INSTRUMENTS	1972		
95	DL A/L		DL A/L	TCPS	HUGHES AIRCRAFT			
96	DL A/L		DL A/L	TCPS	HUGHES AIRCRAFT			
97	DL A/L		DL A/L	TCPS	HUGHES AIRCRAFT			

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 Table D-3. Systems Background Raw Data  
 Part 4: PERSONNEL

ID	Site	System	Software System								TOTAL	LEAST ELEV	SHLL # ELEV2	# ELEV3	SHLL # ELEV4	MOST # ELEV % TIME
			L1	L2	L3	L4	L5	L6	L7	L8						
1	NWHD	LSB	CSS								100	100	100	100	100	
2	NKHD	MEBU									100	100	100	100	100	
3	NWHD	NCS									100	100	100	100	100	
4	NWHD	SSL									100	100	100	100	100	
5	WR-AIC	ALR-46									50	50	50	50	50	
6	WR-AIC	ALR-67									70	70	70	70	70	
7	WR-AIC	ANALD	131								40	40	40	40	40	
8	WR-AIC	AN/AI/L	131								60	60	60	60	60	
9	WR-AIC	AN/AI/L	131								50	50	50	50	50	
10	WR-AIC	APR-38									110	110	110	110	110	
11	WR-AIC	B-52	EVS ATE								75	75	75	75	75	
12	WR-AIC	E SA AVIONICS ATE									25	25	25	25	25	
13	WR-AIC	E 3A AVIONICS ATE									10	10	10	10	10	
14	WR-AIC	F 15									100	100	100	100	100	
15	WR-AIC	F 15									11	11	11	11	11	
16	WR-AIC	F 15, AVIONICS ATE									50	50	50	50	50	
17	WR-AIC	J11Ds									10	10	10	10	10	
18	WR-AIC	J11Ds									50	50	50	50	50	
19	WR-AIC	J11Us									100	100	100	100	100	
20	WR-AIC	J11Us									14	14	14	14	14	
21	WR-AIC	FAVE TAII									15	15	15	15	15	
22	WR-AIC	FAVE TAII									10	10	10	10	10	
23	WR-AIC	F 111D									50	50	50	50	50	
24	WR-AIC	F 111F									10	10	10	10	10	
25	WR-AIC	Fb-111Es									50	50	50	50	50	
26	CASTLE AFB	B-52									40	40	40	40	40	
27	CASTLE AFB	B-52									100	100	100	100	100	
28	CASTLE AFB	E/C-125									10	10	10	10	10	
29	CASTLE AFB	I-4 TRAININER									43	43	43	43	43	
30	UD-AIC	F-16									80	80	80	80	80	
31	UD-AIC	F 16									100	100	100	100	100	
32	UD-AIC	F 16									90	90	90	90	90	
33	UD-AIC	F 16									70	70	70	70	70	
34	UD-AIC	F 16									85	85	85	85	85	
35	UD-AIC	F 16									100	100	100	100	100	
36	UD-AIC	F 16									100	100	100	100	100	
37	UD-AIC	F 4E									5	5	5	5	5	
38	UD-AIC	F 4E									100	100	100	100	100	
39	UD-AIC	F 4E									50	50	50	50	50	
40	UD-AIC	F 4E									70	70	70	70	70	
41	UD-AIC	F 4E									100	100	100	100	100	
42	UD-AIC	F 4E									100	100	100	100	100	
43	UD-AIC	F 4E									100	100	100	100	100	
44	UD-AIC	F 4E									100	100	100	100	100	
45	UD-AIC	F 4E									100	100	100	100	100	
46	UD-AIC	F 4E									100	100	100	100	100	
47	UD-AIC	F 4E									100	100	100	100	100	
48	UD-AIC	F 4E									100	100	100	100	100	
49	UD-AIC	F 4E									100	100	100	100	100	
50	UD-AIC	F 4E									100	100	100	100	100	
											100	100	100	100	100	

Table D-5. Systems Background Raw Data  
Part 4: PERSONNEL

LU	SITE	SYSTEM	SOFTWARE SYSTEM			TOTAL	LEAST #LEV1	GRL1 #LEV2	MST #LEV3	#LEV4	MST % TIME
			MINUTEMAN	MINUTEMAN	MINUTEMAN						
51	UD-HL	F-46	WING-1/ALM			100	3	8	4	5	
51	UD-HL	MINUTEMAN	WING-1/HS-29			50	3	8	30		
51	UD-HL	MINUTEMAN	WINGS/HIS-28			50	1	1	1	25	
54	UD-HL	MINUTEMAN	SSPS/CAPS			25	1	1	1	25	
55	UD-HL	MINUTEMAN	WING-V/HES/KATS			10	1	1	1	10	
56	UD-HL	MINUTEMAN	WING-V/HES/KATS			10	1	1	1	10	
57	UD-HL	RF-4C	AN/ARN-101			30	1	1	1	1	
58	UD-HL	RF-4C	AN/ARN-101			50	1	1	1	50	
59	UD-HL	RF-4C	AN/ARN-101			30	1	1	1	30	
61	UD-HL	RF-4C	AN/ARN-101			20	1	1	1	20	
61	UD-HL	RF-4C	AN/ARN-101			20	1	1	1	20	
61	UD-HL	RF-4C	AN/ARN-101			40	1	1	1	40	
61	UD-HL	RF-4C	AN/ARN-101			40	1	1	1	40	
62	UD-HL	RF-4C	L11			40	1	1	1	40	
64	UD-HL	RF-4C	LFT			40	1	1	1	40	
65	UD-HL	RF-4C	LFT			72	1	1	1	72	
66	UD-HL	RF-4C	ORF			72	1	1	1	72	
67	UD-HL	RF-4C	CADL			10	1	1	1	10	
68	UD-HL	RF-4C	C11S			70	1	1	1	70	
69	UD-HL	RF-4C	ENH			10	1	1	1	10	
70	UD-HL	RF-4C	F/GMS			10	1	1	1	10	
71	UD-HL	RF-4C	RNS			5	1	1	1	5	
72	UD-HL	RF-4C	URS			70	1	1	1	70	
73	UD-HL	RF-4C	HN51			20	1	1	1	20	
74	UD-HL	RF-4C	FTBS			90	1	1	1	90	
75	UD-HL	RF-4C	MIC-1 EXEC			15	1	1	1	15	
76	UD-HL	RF-4C	MIC-2 EXEC			47	1	1	1	47	
77	UD-HL	RF-4C	INS			10	1	1	1	10	
78	UD-HL	RF-4C	ORIGA			60	1	1	1	60	
79	UD-HL	RF-4C	SMCF			20	1	1	1	20	
80	UD-HL	RF-4C	SECF			75	1	1	1	75	
81	UD-HL	RF-4C	SKS/CP			50	1	1	1	50	
82	UD-HL	RF-4C	DS-S			80	1	1	1	80	
83	UD-HL	RF-4C	MDID			100	1	1	1	100	
84	UD-HL	RF-4C	HW-1			50	1	1	1	50	
85	UD-HL	RF-4C	UF-F			100	1	1	1	100	
86	UD-HL	RF-4C	WCS			50	1	1	1	50	
87	UD-HL	RF-4C	UF-F			100	1	1	1	100	
88	UD-HL	RF-4C	HW-F			50	1	1	1	50	
89	UD-HL	RF-4C	UTILITIES			80	1	1	1	80	
90	UD-HL	RF-4C	ASST/UF-UCH			100	1	1	1	100	
91	UD-HL	RF-4C	SFRS			50	1	1	1	50	
92	UD-HL	RF-4C	TCS			100	1	1	1	100	
93	UD-HL	RF-4C	UF-F			100	1	1	1	100	
94	UD-HL	RF-4C	UF-F			100	1	1	1	100	
95	UD-HL	RF-4C	UF-F			100	1	1	1	100	
96	UD-HL	RF-4C	UF-F			100	1	1	1	100	
97	UD-HL	RF-4C	UF-F			100	1	1	1	100	
98	UD-HL	RF-4C	UF-F			100	1	1	1	100	
99	UD-HL	RF-4C	UF-F			100	1	1	1	100	
100	UD-HL	RF-4C	UF-F			100	1	1	1	100	

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Table D-3. Systems Background Raw Data  
Part 3. SUPPORT SYSTEMS(PARTIAL)

ID	SITE	SYSTEM	SUPPORT COMPUTER SYSTEM	2. TIME DEULATED
1	ND-KD	1.55 ME-BU	1.55 ME-BU	50
2	ND-KD	NCS	NCS	0
3	ND-KD	SGC	SGC	0
4	ND-KD	ALR 40	ALR 40	50
5	WR-ALC	ALR 64	ALR 64	0
6	WR-ALL	ALR 64	ALR 64	0
7	WR-ALC	AN/ALQ-151	AN/ALQ-151	0
8	WR-ALC	AN/ALQ-154	AN/ALQ-154	0
9	WR-ALL	AN/ALU 1.11	1.11	0
10	WR-ALC	HP-38	HP-38	0
11	WR-ALC	HP-52; EVG A/E	HP-52; EVG A/E	0
12	WR-ALL	E-5A AVIONICS A/E	AN/GSM 285(B)	0
13	WR-ALL	E-7A AVIONICS A/E	AN/GSM 285(W)	0
14	WR-ALL	E-15	E-15	0
15	WR-ALC	E-15	KADAK	0
16	WR-ALC	E-15 AVIONICS A/E	ADIS, AIS ASIT/OLC	0
17	WR-ALC	J110s	J110s	0
18	WR-ALL	J110s	J110s	0
19	WR-ALC	J110s	J110s	0
20	WR-ALC	J110s	J110s	0
21	WR-ALC	JAVE (W)	JAVE (W)	100
22	WR-ALC	JAVE (A)	JAVE (A)	100
23	SM-ALC	J1110	J1110	40
24	SM-ALL	F 1111	WNL	0
25	SM-ALC	I E 1111	I E 1111	0

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Table D-5. Systems Background Raw Data  
Part 5: SUPPORT SYSTEMS (PARTIAL)

LU	SITE	SYSTEM	SUPPORT COMPUTER SYSTEMS	% TIME DEDICATED
~6	LA SITE AFB	B-52	C/F-I	10
~7	LA SITE AFB	B-52	WSI	0
~8	LASTE AFB	F/C-135 <sup>C</sup>	WSI	0
~9	LASTE AFB	I-4 TRAINER	T-4 SIMULATOR	0
10	DO MIL	F-16	VCC	100
31	DO AIL	F-16	HUD	80
32	DO AIL	F-16	UF-I	85
33	DO AIL	F-16	FLR	100
34	DO AIL	F-16	SMS	30
35	DO AIL	F-16	VAX 11/750, DEC 10, RAINBOW 100, GEE ALSO FEFP, CKSF, AND OS/CHF	60
36	DO AIL	F-4	MOTS S/W DEV. SYSTEM, MOTS FIELD SYSTEM, STATIC SIMULATOR TEST STAND, DYNAMIC TEST STAND	80
36	DO MIL	F-4	MOTS S/W DEV. SYSTEM, MOTS FIELD SYSTEM, STATIC SIMULATOR TEST STAND, DYNAMIC TEST STAND	60
37	DO MIL	F-4E	AN/ARN-101	60
38	DO MIL	F-4E	AN/ARN-101	50
39	DO MIL	F-4E	DEC VAX 785, PDP 11/34, IBM 3083, SEL 32/75, HF 10/10	45
40	DO MIL	F-4E	DEC VAX 785, PDP 11/34, IBM 3083, SEL 32/75, HF 10/10	45
41	DO MIL	F-4E	DEC VAX 785, PDP 11/34, IBM 3083, SEL 32/75, HF 10/10	45
42	DO MIL	F-4E	DEC VAX 785, PDP 11/34, IBM 3083, SEL 32/75, HF 10/10	45
43	DO MIL	F-4E	DEC VAX 785, PDP 11/34, IBM 3083, PDP 11/34, + DF 11/60, SEL	10
44	DO AIL	F-4E	VAX 782, VAX 785, IBM 3083, PDP 11/34, + DF 11/60, SEL	10
45	DO MIL	F-4E	VAX 782, VAX 785, IBM 3083, PDP 11/34, + DF 11/60, SEL	10
46	DO MIL	F-4E	VAX 782, VAX 785, IBM 3083, PUP 11/34, + DF 11/60, SEL	10
47	DO MIL	F-4E	VAX 782, VAX 785, IBM 3083, PDP 11/34, + DF 11/60, SEL	10
48	DO MIL	F-4E	DEC 1160, SEL 32/75, VAX	40
49	DO MIL	F-4E	PDP 1160, SEL 32/75, VAX	40
50	DO MIL	F-4E	PDP 1160, SEL 32/75, VAX	40
51	DO MIL	F-4E	FET IN ELMER 7/32, 6 TERMINALS	40
52	DO MIL	F-4E	FET IN ELMER 7/32, 6 TERMINAL	10
53	DO MIL	F-4E	FET IN ELMER 7/32, 6 TERMINAL	40

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Table D-5. Systems Background Raw Data  
Part b: SUPPORT SYSTEMS(FPARTIAL)

ID	Site	System	Software System		Support Computer Systems		% TIME DEDICATED
			% TIME	DEDICATED	% TIME	DEDICATED	
5.4	DO ALC	MINUTEMAN II BASIS CAPS	IBM 4341		0		0
5.5	DO ALC	MINUTEMAN II WING V/HED/KATS	DEC VAX 785, PDP 11/54, IBM 3083, SEL, PDP-11/60		0		0
5.6	DO ALC	MINUTEMAN II	DEC VAX 785, PDP 11/54, IBM 3083, SEL, PDP-11/60		0		0
5.7	DO ALC	KF-4C	IBM 3083 (4 TERMINALS), VAX 11/785 (UNLIMITED TERMINALS)		20		40
5.8	DO ALC	RF-4C	DEC VAX 785, PDP 11/54, IBM 3083, SEL 3275, HF-1010		45		45
5.9	DO ALC	KF-4C	DEC VAX 785, PDP 11/54, IBM 3083, SEL 3275, HF-1010		90		90
6.0	DO ALC	KF-4C	DEC VAX 785, PDP 11/54, IBM 3083, SEL, PDP-11/60, VAX 11/785		90		90
6.1	DO ALC	KF-4C	DEC VAX 785, PDP 11/54, IBM 3083, SEL, PDP-11/60, VAX 11/785, PDP 11/54, IBM 3083, SEL, PDP-11/60, VAX 11/785		90		90
6.2	DO ALC	KF-4C	DEC VAX 785, PDP 11/54, IBM 3083, SEL, PDP-11/60, VAX 11/785		90		90
6.3	DO ALC	KF-4C	DEC VAX 785, PDP 11/54, IBM 3083, SEL, PDP-11/60, VAX 11/785		90		90
6.4	DO ALC	KF-4C	IBM 4341, ELECTRONIC SYSTEMS TEST SET (TESTS), TRANSLATE EDIT SOFTWARE STATION (TESTS)		90		90
6.5	DO ALC	KF-4C	IBM 4341, INSTRUCTION LEVEL SIMULATOR, SUBSYSTEM SIMULATOR		90		90
6.6	DO ALC	KF-4C	IBM 4341, INSTRUCTION LEVEL SIMULATOR, SUBSYSTEM SIMULATOR		90		90
6.7	DO ALC	b-1B	VAX 11/780, IBM 4341		1		1
6.8	DO ALC	b-1B	VAX 11/780, IBM 4341		1		1
6.9	DO ALC	b-1B	VAX 11/780, IBM 4341		1		1
7.0	DO ALC	b-1B	VAX 11/780, IBM 4341		1		1
7.1	DO ALC	b-1B	VAX 11/780, IBM 4341		1		1
7.2	DO ALC	b-1B	VAX 11/780, IBM 4341		1		1
7.3	DO ALC	b-52	DEC 11/23, 2 FLUO DISKS, FERIN-ELMER 3240, 2 300-MBYTE DISKS,		100		100
7.4	DO ALC	b-52	9-TRACK TAPE DRIVES, 2 KUM MSE / 4 COMPUTERS		95		95
7.5	DO ALC	b-5-	IBM 4341, HARRIS 500, CARD READER, TERMINALS, ETC.		5		5
7.6	DO ALC	E-2	IBM 4341, AMIGA 400, VAX 11/785		25		25
7.7	DO ALC	E-2	HP 2111F, IBM 2111F		0		0
7.8	DO ALC	E-2	IBM 4341, E-3 AITSF		1		1
7.9	DO ALC	E-2	E-3 KADOR SYSTEMS (LURK AND MURKIN), IBM 4341		1		1
8.0	DO ALC	E-2	IBM 4341		1		1
B-	DO ALC	DO-S	KUM 1666B, ENHANCED DISK PRODUCTION SYSTEM, 2 DUAL KASS DRIVES, DATA GENERAL MV1000,		0		0
C-	DO ALC	DO-CM	KUM 1666D, KASS DRIVE, ZEBRA DISK DRIVE, SURFACE DEVELOPMENT SYSTEM		45		45
D-	DO ALC	DO-T	VAX 11/785		0		0
E-	DO ALC	DO-T	KUM 1666B		0		0
F-	DO ALC	DO-T	VAX 11/785, ARRIVAL, IBM 4341, 16001, HONEYwell, NCR 1010, DYNACOM 1010		0		0

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Table D-5. Systems Background Raw Data  
Part 5: SUPPORT SYSTEMS (PARTIAL)

ID	Site	System	% TIME DEDICATED	
			SUPPORT SYSTEM	SUPPORT COMPUTER SYSTEMS
BB	WINDSOR H/B	E-3A	AUCF	IBM 370/168, 3 SPECIAL PURPOSE IBM-4PI SIMULATORS
BY	WINDSOR H/B	E-3A	UTILITIES	IBM 370/168, 3 SPECIAL PURPOSE IBM-4PI SIMULATORS
Y1	LANGLEY	J1105	ASIT/1POCF	IBM 4341 WITH 4 MEG. MAIN MEMORY IN 1M BLOCKS ON LINE STORAGE, FRINICH, 570/0 TERMINALS, INTERFACE SIMULATOR ANALYZER (ISA) NOVA 4/X (INCLUDES 25M DISK, PRINTER, TERMINAL) IBM 360/70, ALL PERIPHERALS, DG-5250, ALL PERIPHERALS, FDF 11/70 TAPE DRIVE FE 3230-1251, TRIDENT DISK DRIVE, REMEX TAPE
Y1	LANGLEY	STRIS	CAFMS	DRIVE, DELTA DATA 7586, CRUMEMCO Z-29 (1) ANALOG MAINFRAME, (4) 16K ROM MINI-COMPUTERS, (5) DC DISK DRIVES SEE CRISP & O'SCAMP, IBM 4341 VS/OS
Y2	LANGLEY	IACS	DC/SR	HUGHES UTIL IBM UTIL
Y3	LANGLEY	LIF-1		IBM 4341
Y4	LANGLEY	LIP-1		H-411B(2), IBM 4341, 408 SIMULATOR, 4Cs KU1
Y5	LANGLEY	407L		55
Y6	LANGLEY	407L		
Y7	LANGLEY	407L		

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07-0-B,Table D-1. Systems Background Raw Data  
Partial SUPPORTABILITY PROBLEMS

ID	SITE	SYSTEM	SOFTWARE SYSTEM	PROBLEM(S) DESCRIPTION
1	NOKAD	CSS MEBU NCS SSC	ALR-46	(1) Lack of tools to locate and debug failures. (1) Lack of tools to locate and debug failures. (1) Inadequate number of trained software personnel. (2) Time to get completed software fielded is much too long.
2	NOKAD			(1) Inadequate number of trained software personnel. (1) Inability to adequately staff positions prior to FMR to ensure software quality.
3	NOKAD			(2) Inability to maintain the required level of expertise. (1) Inability to adequately staff positions prior to FMR to ensure software quality.
4	NOKAD			(2) Inability to maintain the required level of expertise. (1) Insufficient trained personnel before and after FMR.
5	WR-ALC	AN/ALU-131	UFF	(2) Inability to maintain required level of expertise.
6	WR-ALC	ALR-6Y		(1) Limited time available on test set where software problems and solutions have to be verified.
7	WR-ALC	AN/ALU-131		(1) Inadequate logistical support in obtaining engineering services, computer, other hardware, consumables, parts, & LRUs.
8	WR-ALC	AN/ALU-131	IUT	(2) Lack of technical personnel in the areas of radar, OFPA, computer science, & hardware.
9	WR-ALC		APR-3B	(3) Lack of understanding of the process and its requirements throughout the AF.
10	WR-ALC	HFR-5B	ASO-151	(1) Configuration control after PRFI. (2) CFIN conversion.
11	WR-ALC	B-51 EVS ATE		(3) Interface between filt. and control.
11	WR-ALC	E SA AVIONICS ATE	AN/GSM-285(B)	
12	WR-ALC	E SA AVIONICS ATE	AN/GSM-285(W)	
13	WR-ALC	F 15	CC	
14	WR-ALC	F 15	KRADAR	
15	WR-ALC	F 15		
16	WR-ALC	F 15 AVIONICS ATE	AUTS, AIP	
17	WR-ALC	J1105	AS11/DCP	
18	WR-ALC	J1105	E-3A AWACS/OLP	
19	WR-ALC	J1105	SF/USEK	
20	WR-ALC	FAVE (ML)	SYS EXEC16EK	
21	WR-ALC	FAVE (ML)	ASIF	
22	WR-ALC	FAVE (ML)	UFF	
23	WR-ALC			WNL
24	WR-ALC			WNC

Table U-3. Systems Background Raw Data  
Part 6: SUPPORTABILITY PROBLEMS

ID	SITE	SYSTEM	SOFTWARE SYSTEM	PROBLEM(S) DESCRIPTION
25	SM-ALC	FB-111A	WNC	(1) Organic software support and configuration management suffer considerably from a shortage and continual turnover of qualified engineers.
26	CASTLE AFB	B-52	CPI	
27	CASTLE AFB	B-52	WSI	
28	CASTLE AFB	I-C-135	WSI	
29	CASTLE AFB	I-4 TRAINER	I-4 SIMULATOR	(1) Number of configurations, customers, change drivers, etc.
30	DO-ALC	F-16	HIC	
31	DO-ALC	F-16	HUO	
32	DO-ALC	F-16	OFI	
33	DU AL	F-16	FLR	
34	UD-ALC	I-16	SMS	(1) Unable to hire sufficient number of technical support people to staff all software projects requested to be addressed during past or present updates.
35	UD ALC	I-4	FDTS	(1) Too much time is required to fully understand and be able to utilize the numerous capabilities of the operating system.
36	UD-ALC	I-4	MUDS	
37	UD-ALC	I-4E	AN/ARN-101	(1) Insufficient manpower.
38	UD-ALC	F-4E	AN/ARN-101	
39	OO-ALC	F-4E	AN/ARN-101	
40	OO-ALC	F-4E	AN/ARN-101	
41	CO-ALC	F-4E	AN/ARN-101	
42	OO-ALC	F-4E	AN/ARN-101	
43	OO-ALC	F-4L	AN/ARN-101	
44	OO-ALC	F-4L	AN/ARN-101	
45	UD-ALC	F-4L	AN/ARN-101	(1) Support system is currently incomplete.
46	UD-ALC	F-4G	AN/ARN-101	(1) Support system is currently incomplete.
47	UD-ALC	F-4L	AN/ARN-101	(1) Support system is currently incomplete.
48	UD-ALC	F-4L	LNU-1/ALM	(1) Most of the work consists of software enhancements that require an engineering background rather than a software background.
49	OO-ALC	F-4L	LNU-1/ALM	
50	OO-ALC	F-4L	LNU-1/ALM	
51	DU AL	MINUTEMAN	WING 11/2015	(1) Lack of engineer's qualified to solve complex algorithmic problems.
52	DU AL	MINUTEMAN	WING V1/HB-29	(1) Lantency of current projects.
53	DU AL	MINUTEMAN	WBNS/HB-29	(1) Lantency of current projects.
54	DU AL	MINUTEMAN	WBNS/CAFS	(1) Outdated software system currently being upgraded.
55	DU AL	MINUTEMAN	WING V1/HB/RHIS	(1) Separate need cycles.
				(2) Separate support functions.

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D-29-005Table D-5. Systems Backup and New Data  
Part 6: SUPPORTABILITY PROBLEMS

ID	SITE	SYSTEM	PROBLEM(S)	DESCRIPTION
56	00-AL*	MINUTEMAN II		
57	00-HLC	KF-4L		
58	00-AL	KF-4C		
59	00-HLC	KF-4L		
60	00-AL	KF-4L		
61	00-AL	KF-4L	(1) Insufficient manpower.	
62	00-HLC	KF-4L		
63	00-HLC	ALCM		
64	00-HLC	ALCM		
65	00-HLC	ALCM		
			(1) A major problem has been the turnover of our experienced people and their replacement with inexperienced personnel.	
			(2) We have problems establishing a training program using our experienced people, since they are committed to other projects.	
			(3) Maintainability of the subsystem simulator is poor.	
66	00-HLL	ALM		
67	00-HLC	LR		
			(1) Inadequate documentation.	
			(2) Extremely poor contractor design.	
			(3) High turnover rate.	
68	00-HLC	K-1B		
			(4) Inadequate documentation.	
			(1) Inadequate manpower.	
			(2) Insufficient manpower.	
			(3) AF/contractor interface.	
69	00-HLC	K-1B		
			(4) High turnover rate.	
			(1) Extremely poor documentation.	
			(2) System is obsolete and will need to be redesigned	
			(3) Matter of years after FMMI.	
			(4) Unique language.	
70	00-HLC	K-1B		
71	00-HLC	K-1B		
			(1) Inadequate trained personnel.	
			(2) Insufficient manpower.	
			(3) Insufficient manpower.	
			(4) Inability to get necessary information from contractor.	
72	00-HLL	K-1B		
			(5) Insufficient trained personnel.	
			(1) The uniqueness of the hardware requires that final development and testing be done on an actual contractor. This complicates development and testing schedules.	
73	00-HLC	LS		
			(1) In the Executive MS document, over 50% of the document does not match the code or training conventions.	
			(1) There is no way to or desire to support the INS program until an inertial computer (INR) is installed in the life cycle support facility (LCS).	
74	00-HLL	b-54		
75	00-HLC	b-54		
76	00-HLL	b-54		
77	00-HLL	b-54		
			(1) In the Executive MS document, over 50% of the document does not match the code or training conventions.	
			(1) There is no way to or desire to support the INS program until an inertial computer (INR) is installed in the life cycle support facility (LCS).	

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Table D-3. Systems Background Raw Data  
Part 2: SUPPORTABILITY PROBLEMS

ID	SITE	SYSTEM	SOFTWARE SYSTEM	PROBLEM(S) DESCRIPTION
78	UK ALIC	E-3A	IMEGA SMIF	(1) NUC 1070 has limited memory. (1) Software configuration management controls. (2) Equipment maintenance problems caused by supply system. (3) Lack of AIFS engineering and configuration management for hardware.
80	UK ALIC	E-3A	SHCP	(1) Substantial overhead work demands. (2) Unavailability of radar and IBM 4341.
81	UK ALIC	E-3A	SKSCH	(1) Flowcharts generated by AUTOFLOW--a system useless for supportability. (2) Software was moved to a less suitable substitute computer.
82	UK ALIC	GLCM	MF-5	(1) Authorized manpower is inadequate. (2) Support equipment is also required for product production.
83	UK ALIC	GLCM	M-DID	(1) Several programs exist which are now being integrated into one consistent package. Without integration, supportability would be more difficult. Because the several programs would require separate maintenance. (1) Documentation is in poor shape.
84	UK ALIC	GLCM	MF-1	(1) Software is implemented on antiquated equipment. (1) Complete dependence on the contractor to support software.
85	UK ALIC	GLCM	MF-5	(1) Poor documentation. (2) Unenforced standards.
86	UK ALIC	GLCM	MF-5	(3) Poor acquisition standards. (4) No coding standards. (5) No delivery standards.
87	UK ALIC	GLCM	MF-5	(1) Poor documentation. (2) Unenforced standards.
88	LINK 44B	E-3A	MF-5	(3) Inadequate testing by contractor.
89	LINK 44B	E-3A	MF-5	(4) Inufficient representation during software design. (1) Lack of HIGHLIGHTS CLASS 1 terminal(s).
90	LINK 44B	E-3A	MF-5	(2) Nonpension of IBM TLR 65 version. (3) Lack of militarized operator interface units(OIUs). (4) No maintenance contract for LS.
91	LINK 44B	E-3A	MF-5	(5) Generally disjointed program management (ESD, IBM, WALKER RUBINS ALIC, etc.). (6) Lack of complete, computer readable documentation. (1) Graphic software is maintained in MIFLU II assembler on non supported operating system. Limited number of knowledgeable operators. (2) No CSD Editor is a line editor and only 1 terminal can be used at a time. (3) Training time is 1 to 18 months.
92	LINK 44B	E-3A	MF-5	(4) Training time is 1 to 18 months.
93	LINK 44B	E-3A	MF-5	(5) Training time is 1 to 18 months.
94	LINK 44B	E-3A	MF-5	(6) MARKS/INTERSTATE
95	LINK 44B	E-3A	MF-5	(7) MARKS/INTERSTATE

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07/26/85Table D-3. Systems Factors Ground Rule Data  
Part 6: SURVIVABILITY REQUIREMENTS

ID	SITE	SYSTEM	DESCRIPTION
46	LANL EY	400/1 100F / 100T	IBM 3270 100F / 100T  (1) No AIC training support in MAP - IBM assembler language. (2) Significant down time on AICs because of age and lack of spare parts. (3) Significant amount of time dedicated to interface testing.

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Table D-4. Systems Evaluation Raw Data  
Part I: SOFTWARE PRODUCT MAINTAINABILITY (AT DELIVERY)

ID SITE	SYSTEM	SOFTWARE SYSTEM	AP DOC MOD	AP DOC DES	AP DOC CON	AP SIM EXP	AP SIM INS	AP SRC CON	AP SRC DES	AP SRC MOD	AP SRC SIM	AP SRC EXP	AP SRC INS	AP SRC PRODUCT	
1 NORAD	CSS	CSS	-10 -20	-20 -20	-20 -20	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10	
2 NORAD	MERU	MERU	20 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99
3 NORAD	NCS	NCS	10 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99	99 -99
4 NORAD	SSC	SSC	10 -15	15 -15	10 -10	5 -5	-10 -5	5 -5	5 -5	10 -10	15 -15	5 -5	10 -10	5 -5	10 -10
5 WR AL C	AL R 46	AL R 46	25 -25	20 -20	23 -23	-5 -29	10 -10	10 -10	10 -10	15 -15	5 -5	5 -5	21 -21	25 -25	20 -20
6 WR AL C	AL K-69	AL K-69	30 -30	20 -20	20 -20	-10 -10	-10 -10	-10 -10	15 -15	5 -5	30 -30	-99 -99	10 -10	15 -15	-10 -10
7 WR AL C	BIG	BIG	-45 -40	-40 -40	-50 -50	10 -10	-50 -50	-40 -40	-30 -30	-40 -40	-40 -40	-40 -40	-40 -40	-40 -40	-40 -40
8 WR AL C	AN/ALQ-131	AN/ALQ-131	-40 -30	-40 -40	-50 -50	10 -10	-50 -50	-70 -70	-50 -50	-20 -20	-40 -40	-50 -50	-40 -40	-40 -40	-40 -40
9 WR AL C	AN/ALQ-131	AN/ALQ-131	-50 -40	-40 -40	-50 -50	-50 -50	-50 -50	-50 -50	-20 -20	-40 -40	-40 -40	-40 -40	-40 -40	-40 -40	-40 -40
10 WR AL C	AN/K-38	AN/K-38	25 -25	15 -20	0 -5	50 -50	50 -50	20 -20	10 -10	50 -50	0 -5	0 -5	0 -5	0 -5	0 -5
11 WR AL C	E-5 EWS ATE	E-5 EWS ATE	40 -45	20 -40	40 -40	55 -55	10 -20	5 -20	1 -20	1 -20	1 -20	1 -20	1 -20	1 -20	1 -20
12 WR AL C	E-7A AVIONICS ATE	E-7A AVIONICS ATE	-20 -20	10 -10	-20 -20	10 -10	10 -10	5 -20	5 -20	5 -20	10 -10	10 -10	10 -10	10 -10	10 -10
13 WR AL C	E-7A AVIONICS ATE	E-7A AVIONICS ATE	30 -30	10 -10	20 -20	25 -25	40 -40	10 -10	40 -40	20 -20	40 -40	10 -10	30 -30	10 -10	30 -30
14 WR AL C	F-15	F-15	25 -20	20 -20	40 -40	10 -10	20 -20	50 -50	30 -30	50 -50	40 -40	40 -40	40 -40	40 -40	40 -40
15 WR AL C	F-15	F-15	40 -40	35 -35	40 -40	50 -50	40 -40	40 -40	50 -50	40 -40	40 -40	50 -50	40 -40	40 -40	40 -40
16 WR AL C	F-15 AVIONICS ATE	F-15 AVIONICS ATE	10 -20	20 -20	40 -40	10 -10	5 -10	5 -10	5 -10	5 -10	5 -10	5 -10	5 -10	5 -10	5 -10
17 WR AL C	JTIDS	JTIDS	35 -30	30 -30	30 -30	35 -35	30 -30	35 -35	30 -30	35 -35	30 -30	30 -30	30 -30	30 -30	30 -30
18 WR AL C	JTIDS	JTIDS	35 -30	30 -30	30 -30	35 -35	30 -30	35 -35	30 -30	35 -35	30 -30	30 -30	30 -30	30 -30	30 -30
19 WR AL C	JTIDS	JTIDS	15 -30	0 -30	-10 -10	-10 -10	0 -10	0 -10	0 -10	0 -10	0 -10	0 -10	0 -10	0 -10	0 -10
20 WR AL C	JTIDS	JTIDS	-15 -20	-20 -20	-10 -10	-10 -10	-20 -20	-20 -20	-10 -10	-20 -20	-10 -10	-20 -20	-10 -10	-20 -20	-10 -10
21 WR AL C	FAVE TACT	FAVE TACT	45 -35	35 -35	40 -40	40 -40	55 -55	45 -45	40 -40	45 -45	40 -40	45 -45	30 -30	45 -45	45 -45
22 WR AL C	FAVE TACT	FAVE TACT	35 -45	40 -40	35 -35	40 -40	35 -35	40 -40	35 -35	40 -40	35 -35	40 -40	35 -35	40 -40	40 -40
23 SM AL C	F-111D	F-111D	20 -25	20 -25	15 -25	15 -25	40 -40	40 -40	40 -40	40 -40	40 -40	10 -20	-10 -10	-10 -10	-20 -20
24 SM AL C	F-111F	F-111F	20 -25	20 -25	20 -25	20 -25	40 -40	40 -40	40 -40	40 -40	40 -40	10 -20	-10 -10	-10 -10	-20 -20
25 SM AL C	FB-111A	FB-111A	20 -25	20 -25	15 -20	15 -20	40 -40	40 -40	40 -40	40 -40	40 -40	10 -20	-10 -10	-10 -10	-20 -20
26 CASTLE	AFB B-52	AFB B-52	5 -20	5 -20	-20 -10	-10 -10	0 -10	0 -10	0 -10	0 -10	0 -10	20 -20	5 -10	-99 -99	5 -10
27 CASTLE	AFB B-52	AFB B-52	5 -20	5 -20	-20 -10	-10 -10	0 -10	0 -10	0 -10	0 -10	0 -10	20 -20	5 -10	-99 -99	5 -10
28 CASTLE	AFB F-1C-175	AFB F-1C-175	5 -5	10 -15	10 -15	-15 -15	-50 -50	5 -5	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10	-10 -10
29 CASTLE	AFB F-14 TRAINER	AFB F-14 TRAINER	20 -25	20 -25	20 -20	20 -20	30 -30	20 -20	30 -30	25 -25	20 -20	20 -20	20 -20	20 -20	20 -20
30 OO AL C	F-16	F-16	15 -30	10 -10	10 -10	5 -5	5 -5	5 -5	5 -5	30 -30	40 -40	10 -10	20 -20	10 -10	10 -10
31 OO AL C	F-16	F-16	5 -5	40 -45	-25 -15	5 -5	40 -45	20 -10	-20 -10	15 -15	40 -40	15 -15	20 -20	10 -10	10 -10
32 OO AL C	F-16	F-16	-30 -10	-40 -40	-30 -10	-20 -20	-25 -10	-20 -20	-15 -10	-40 -40	-30 -30	-20 -20	-20 -20	-20 -20	-20 -20
33 OO AL C	F-16	F-16	-99 -10	20 -20	10 -20	20 -20	18 -18	10 -20	20 -20	18 -18	10 -20	10 -20	10 -20	10 -20	10 -20
34 OO AL C	F-4E	F-4E	25 -35	20 -20	10 -10	20 -20	25 -25	20 -20	25 -25	20 -20	30 -30	10 -10	25 -25	10 -10	10 -10
35 OO AL C	F-4E	F-4E	10 -30	10 -10	30 -10	10 -10	30 -20	10 -10	30 -20	10 -10	30 -30	10 -10	20 -20	10 -10	10 -10
36 OO AL C	F-4E	F-4E	-10 -10	10 -20	10 -20	20 -20	25 -10	10 -10	10 -10	10 -10	10 -10	-10 -10	10 -10	10 -10	10 -10
37 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
38 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
39 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
40 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
41 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
42 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
43 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
44 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
45 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40
46 OO AL C	F-4E	F-4E	40 -50	30 -35	40 -35	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40	30 -35	40 -40

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Table D-4. Systems Evaluation Raw Data  
Part II: SOFTWARE PRODUCT MAINTAINABILITY (AT DELIVERY)

ID SITE	SYSTEM	SOFTWARE SYSTEM	Scale: 50 (Low) to + 99 (High) + 99 (N/A)											
			AP DOC MOD DES	AP DOC MOD DES	AP SRC SIM CON									
47 00 ALC	F-46	AN/ARN-101	25	30	-5	-5	-5	-5	-5	-5	-5	-5	-5	-5
48 00 ALC	F-46	LRU-1/ACM	20	20	50	10	10	20	30	20	30	10	10	20
49 00 ALC	F-46	LRU-1/ACM	-10	45	30	40	30	40	30	40	40	40	40	35
50 00 ALC	F-46	LRU-1/ACM	-25	25	15	25	25	40	10	20	25	10	25	10
51 00 ALC	MINUTEMAN	WING 11/2015	15	25	5	5	5	25	15	25	5	10	5	5
52 00 ALC	MINUTEMAN	WING V1/HG-29	99	99	99	99	99	99	99	99	99	99	99	99
53 00 ALC	MINUTEMAN	WINGS/HG-28	25	25	10	35	20	30	10	25	20	30	10	25
54 00 ALC	MINUTEMAN	SSAS/CAPS	40	40	20	20	20	20	-99	40	40	40	-99	40
55 00 ALC	MINUTEMAN	WING V1/HG/RATS	40	30	40	20	-20	-50	-99	40	40	50	-50	-99
56 00 ALC	MINUTEMAN	WING V1/HEG/RATS	40	40	50	20	-30	-99	40	40	50	-30	-50	-99
57 00 ALC	KF-4C	AN/ARN-101	20	45	10	20	25	-10	15	40	10	20	-10	17
58 00 ALC	KF-4C	AN/ARN-101	40	50	20	25	30	-50	10	25	20	-10	10	30
59 00 ALC	KF-4C	AN/ARN-101	5	20	-15	30	-40	20	-10	-15	20	-10	-45	-20
60 00 ALC	KF-4C	AN/ARN-101	5	40	5	20	15	20	15	40	30	10	5	20
61 00 ALC	KF-4C	AN/ARN-101	-10	30	-10	20	-20	-30	30	10	30	-20	-10	10
62 00 ALC	LIT	AN/ARN-101	-10	10	-20	10	10	10	10	10	30	-10	10	5
63 00 ALC	ALCM	LIT	40	50	40	40	40	40	30	40	50	45	50	40
64 00 ALC	ALCM	OF P	35	40	40	30	40	35	40	30	40	30	30	30
65 00 ALC	ALCM	OF F	-20	25	-40	15	-40	-10	25	30	35	35	20	0
66 00 ALC	ALCM	CADC	-25	-25	-20	-15	-10	-40	-40	-40	-50	-40	-50	-30
67 00 ALC	B-1B	CITS	-30	-99	-99	-99	-99	-99	-99	-99	-28	-99	-99	-29
68 00 ALC	B-1B	EMUX	21	-99	-99	-99	-99	-99	-99	-40	-99	-99	-99	-13
69 00 ALC	B-1B	F/LGMS	-35	-99	-99	-99	-99	-99	-99	-99	2	-99	-99	-35
70 00 ALC	B-1B	INS	-15	-99	-99	-99	-99	-99	-99	-99	10	-99	-99	-10
71 00 ALC	B-1B	ORS	10	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-25
72 00 ALC	B-1B	BRIS	-50	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-50
73 00 ALC	B-1B	FTSS	25	25	20	20	30	-5	35	40	10	35	20	40
74 00 ALC	B-1B	MC-1 EXEC	15	30	5	-5	10	20	30	40	10	10	1	20
75 00 ALC	B-1B	MC-2 EXEC	40	45	45	40	45	45	45	45	40	-20	10	38
76 00 ALC	B-1B	INS	-40	25	5	-10	20	5	-20	-20	-20	-20	-45	40
77 00 ALC	E-1A	OMEGA	10	15	10	5	10	10	20	10	10	10	20	10
78 00 ALC	E-1A	SMCP	-40	99	99	99	99	99	-10	25	-10	-10	20	10
79 00 ALC	E-1A	SKCF	-25	15	-40	5	40	5	99	20	15	20	5	99
80 00 ALC	E-1A	SKFCF	-15	1	-5	-5	-5	-1	-10	1	-5	-5	-10	-5
81 00 ALC	E-1A	DR-S	15	25	10	25	25	-45	25	25	35	35	35	35
82 00 ALC	E-1A	GLCM	40	45	45	35	40	35	30	30	30	30	30	35
83 00 ALC	E-1A	M-DID	20	20	20	20	20	20	20	20	15	15	20	20
84 00 ALC	GLCM	RF-T	-10	-20	-15	10	-10	10	-10	25	-15	-10	-15	10
85 00 ALC	GLCM	WE-S	25	30	25	15	-15	-15	-15	15	-15	-15	-15	15
86 00 ALC	GLCM	WF-P	-15	-20	-30	-15	-15	-15	-15	-15	-10	-10	-10	-10
87 00 ALC	SKAM	WF-F	25	-10	-20	15	-20	-20	-20	-10	10	-10	-50	10
88 00 ALC	E-1A	WFL	-30	-20	-40	-30	-40	-50	-50	-40	-40	-30	-30	-30
89 100 ALC	E-1A	WFL WFL	-30	-30	-40	-20	-20	-20	-20	-20	-20	-20	-20	-20
90 100 ALC	E-1A	J11DS	25	30	25	20	25	25	30	30	30	25	25	25
91 100 ALC	E-1A	SIRTS	25	25	25	10	25	-10	15	25	-25	10	25	25
92 100 ALC	E-1A	TALS	40	20	19	10	50	10	10	10	10	20	50	20

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Table D-4. Systems Evaluation Raw Data  
Part I: SOFTWARE PRODUCT MAINTAINABILITY (AT DELIVERY)

Scale: - 50 (Low) to + 50 (High) 1 + 99 (Missing) 1 + 99 (N/A)

ID SITE	SYSTEM	AP			AP													
		DOL	DOC	DOC	DOC	DOC	SRC	PRO-	PRO-									
	SYSTEM	MOD	DES	CON	BIM	EXP	INS	MOD	DES	CON	BIM	EXP	INS	MOD	DES	CON	DUCT	DUCT
93 LANGLEY	TIP2	-30	-10	-50	-50	-50	10	-10	-40	-20	-50	-50	-50	-20	-50	-50	-40	-40
94 LANGLEY	TIP1	-30	30	-10	30	10	-40	25	30	20	20	-40	-40	-25	-40	-40	0	0
95 LANGLEY	HUGHES UTIL	5	10	10	10	5	5	0	-30	-10	-15	-20	-30	-40	-10	-10	-20	-20
96 LANGLEY	IBM UTIL	30	40	30	40	40	40	30	30	40	40	40	40	40	40	40	20	20
97 LANGLEY	10NF/1MFP	10	40	10	10	5	40	35	-10	10	-20	10	-10	25	10	25	35	10

Table D-4. Systems Evaluation Raw Data

Part 2: SOFTWARE SUPPORT FACILITY (AT DELIVERY)

ID SITE			SYSTEM			Scales: -50(Low) to +50(High) 1 - 99(Missing) 1 + 99(N/A)														
						AE	AE	AE	AE	AE	AE	AE	AE	AE	AE	AE	AE	AE	AE	
						PER MAN	PER TEC	PER SUP	PER CON	PER HOS	PER LAB	SYS BEN	SYS OPE	SYS 0TH	SYS OTH	SYS BEN	SYS OPE	SYS 0TH	FAC OFF	FAC ENV
1	NORAD	CSS	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-25
2	NORAD	MEBU	-10	10	-10	99	-99	-99	-20	99	99	99	-50	99	-10	-10	-10	-10	-10	-50
3	NORAD	NCS	-99	10	10	-10	99	-99	-20	99	99	99	-20	99	-25	25	10	-10	-10	-50
4	NORAD	SSC	25	25	30	20	30	-10	10	99	99	-20	99	-25	25	10	-10	-10	-10	-50
5	WR-ALC	ALR-46	30	10	20	15	5	-20	-30	-10	-10	10	99	-5	10	10	-10	-10	-10	-50
6	WR-ALC	ALR-69	-20	10	10	15	5	-20	20	20	20	10	99	20	20	20	10	10	10	-10
7	WR-ALC	AN/ALQ-131	-40	10	40	-40	10	-40	10	-40	10	-40	-30	10	-40	-30	-30	-30	-30	-40
8	WR-ALC	AN/ALQ-131	-30	10	-40	-45	10	-40	-30	-35	-40	-30	-35	-40	-30	-30	-30	-30	-30	-50
9	WR-ALC	AN/ALQ-131	-50	10	-40	-50	-10	-30	-30	-30	-40	-30	-30	-40	-30	-30	-30	-30	-30	-40
10	WR-ALC	APR-3B	-25	25	0	20	25	20	0	25	25	25	25	25	25	25	25	25	25	25
11	WR-ALC	B-52 EVS ATE	35	40	40	20	30	30	30	10	99	20	99	30	30	30	30	30	30	30
12	WR-ALC	E-7A AVIONICS ATE	40	25	40	40	99	25	20	5	45	20	99	35	10	35	40	30	30	30
13	WR-ALC	E-3A AVIONICS ATE	40	25	40	40	99	25	20	5	45	20	99	35	10	35	40	30	30	30
14	WR-ALC	F-15	20	10	-20	20	10	40	40	30	40	40	99	30	40	30	40	30	40	20
15	WR-ALC	F-15	10	5	35	45	25	25	25	25	25	25	25	25	25	25	25	25	25	25
16	WR-ALC	F-15 AVIONICS ATE	-20	10	-20	20	10	-20	20	10	-10	10	-5	20	-5	10	99	20	20	20
17	WR-ALC	JTIDS	20	25	29	15	10	15	10	15	10	15	10	5	0	20	-20	30	30	25
18	WR-ALC	JTIDS	25	20	20	15	10	15	10	15	10	15	10	5	20	-10	30	30	30	25
19	WR-ALC	JTIDS	35	-99	15	-99	99	-5	99	-5	99	-5	99	99	5	5	5	5	5	-99
20	WR-ALC	JTIDS	25	99	25	99	-20	-20	99	99	99	99	99	99	0	0	0	0	0	0
21	WR-ALC	F-AVE TACK	40	40	40	40	40	40	45	45	45	35	35	35	40	40	40	40	40	40
22	WR-ALC	F-AVE TACK	30	35	38	38	40	38	35	40	40	40	40	40	40	40	40	40	40	35
23	SM-ALC	F-111D	40	30	40	30	30	20	20	30	20	30	20	30	20	20	20	20	20	25
24	SM-ALC	F-111F	40	30	40	30	30	20	20	30	20	30	20	30	20	20	20	20	20	20
25	SM-ALC	FB-111A	40	30	40	30	30	20	20	30	20	30	20	30	20	20	20	20	20	15
26	CASTLE	AFB B-52	-30	-30	-30	-30	30	5	5	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-20
27	CASTLE	AFB B-52	10	20	10	10	10	20	10	20	10	20	10	5	5	10	5	10	5	-20
28	CASTLE	AFB F-1C-135	10	20	10	10	10	20	10	10	10	10	10	10	0	20	-20	-20	-20	-20
29	CASTLE	AFB T-4 TRAINER	5	5	10	5	-20	10	10	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-10
30	DO-ALC	F-16	0	0	-10	0	20	30	30	30	30	30	30	30	30	30	-99	-99	-99	-99
31	DO-ALC	F-16	-15	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-30	-30	-30	-10
32	DO-ALC	F-16	10	-30	30	10	99	45	40	99	99	50	20	45	45	45	50	50	50	50
33	DO-ALC	F-16	-40	0	-30	-40	-50	-30	-20	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	-35
34	DO-ALC	F-16	18	10	20	10	30	8	-10	0	20	20	10	5	10	5	10	5	10	
35	DO-ALC	F-4	30	40	40	25	25	20	20	20	20	20	15	-20	-20	-20	-20	-20	-20	-25
36	DO-ALC	F-4	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
37	DO-ALC	F-4E	28	40	15	45	28	25	25	25	25	25	25	25	25	25	25	25	25	25
38	DO-ALC	F-4E	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40
39	DO-ALC	F-4E	30	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40
40	DO-ALC	F-4E	40	40	45	40	20	10	10	-20	99	10	-20	99	10	-20	99	10	10	10
41	DO-ALC	F-4E	10	-20	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10
42	DO-ALC	F-4E	30	20	30	40	-10	-20	99	-10	10	-10	-10	-10	-10	-10	-10	-10	-10	-10
43	DO-ALC	F-4E	25	20	-5	20	25	-10	-10	-10	-10	-10	-10	-10	-10	-10	-10	-10	-10	-10
44	DO-ALC	F-4E	20	25	5	20	20	20	20	20	20	20	20	20	20	20	20	20	20	20
45	DO-ALC	F-4E	20	20	5	20	20	20	20	20	20	20	20	20	20	20	20	20	20	20
46	DO-ALC	F-4E	20	20	-5	20	20	20	20	20	20	20	20	20	20	20	20	20	20	20

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Table D-4. Systems Evaluation Raw Data  
Part 2: SOFTWARE SUPPORT FACILITY (AT DELIVERY)

ID	SITE	SYSTEM	Scale: -5 (LOW) to +5 (HIGH)						- 99 (Missing) + 99 (N/A)					
			AE	AE	AE	AE	AE	AE	SYS	SYS	SYS	FAC	FAC	ENV
			PER	PER	PER	PER	CON	HDS	BEN	LAB	OTH	OFF	ENV	IRON
F-4G	F-4C	AN/ARN-101	20	-5	20	-10	-10	-99	99	-99	-99	35	35	30
F-4G	F-4C	LRU-1/ACM	20	20	10	10	10	-20	-99	10	10	10	10	10
F-4G	F-4C	LRU-1/ACH	25	25	25	99	30	35	25	-10	30	30	25	25
F-4G	F-4C	LRU-1/ACH	25	25	20	99	10	15	15	15	-99	20	20	20
MINUTEMAN	MINUTEMAN	WING 11/20/15	40	25	45	45	5	5	5	5	-99	-40	35	-5
MINUTEMAN	MINUTEMAN	WING VI/HB-24	45	25	45	45	-5	-5	-5	-5	-5	-5	-5	-5
MINUTEMAN	MINUTEMAN	WINGS/HS-28	40	25	45	45	40	40	40	40	-99	-40	40	-5
MINUTEMAN	MINUTEMAN	SSAS/CAFS	40	40	40	40	50	50	40	40	-99	40	40	40
MINUTEMAN	MINUTEMAN	WING V/HEG/KATS	20	20	30	30	45	45	20	40	-99	40	40	30
MINUTEMAN	MINUTEMAN	WING VI/HEG/KATS	40	30	30	30	40	30	40	40	-99	40	40	40
AN/ARN-101	AN/ARN-101	AN/ARN-101	28	40	15	45	28	25	99	15	15	-99	40	40
AN/ARN-101	AN/ARN-101	AN/ARN-101	40	40	40	40	40	20	40	40	40	40	40	40
AN/ARN-101	AN/ARN-101	AN/ARN-101	30	40	40	40	40	-20	-25	-15	-99	-15	-15	-5
AN/ARN-101	AN/ARN-101	AN/ARN-101	40	40	40	40	40	20	-5	5	5	-99	-5	-5
AN/ARN-101	AN/ARN-101	AN/ARN-101	10	-20	10	10	10	-20	99	10	10	-99	20	10
AN/ARN-101	AN/ARN-101	AN/ARN-101	30	20	30	30	40	-10	99	-10	10	-99	-30	-10
LIT	LIT	ALCM	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50
LPT	LPT	ALCM	15	10	5	40	-50	-50	-50	-50	-50	-50	-50	-50
UF P	UF P	ALCM	15	5	15	0	30	-50	-50	-50	-50	-50	-50	-50
UF P	UF P	ALCM	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50
CADC	CADC	ALCM	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
CITS	CITS	ALCM	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
EMUX	EMUX	ALCM	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
F/ICMHS	F/ICMHS	ALCM	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
INS	INS	ALCM	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
ORS	ORS	ALCM	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
BNSI	BNSI	ALCM	15	40	35	30	10	30	45	5	15	30	99	40
F155	F155	ALCM	35	40	30	35	40	47	50	50	50	99	50	45
MC-1 EXEC	MC-1 EXEC	ALCM	20	10	15	20	10	20	10	10	20	99	20	20
MC-2 EXEC	MC-2 EXEC	ALCM	35	99	35	10	99	20	35	99	20	99	10	45
LNS	LNS	E-3A	10	10	-10	-10	30	-30	10	-10	-50	30	99	45
ONEGA	ONEGA	E-3A	-10	-10	-10	-20	-20	-25	-30	-99	-40	-10	-30	-30
SIMCP	SIMCP	E-3A	-20	-30	5	-20	10	10	10	10	10	99	-10	-20
SKCP	SKCP	E-3A	5	-10	30	-15	5	10	10	-10	-15	99	-5	-10
SKGSLP	SKGSLP	E-3A	15	20	25	-10	15	45	45	99	45	99	15	5
DFS	DFS	E-3A	-35	-40	-40	-30	40	10	40	10	-10	20	25	30
M-DID	M-DID	E-3A	30	20	30	30	40	30	30	40	30	99	40	30
MFT	MFT	E-3A	10	-20	10	-10	15	30	35	30	40	99	20	20
GLCM	GLCM	E-3A	84	90	90	90	90	90	90	90	90	90	90	90
GLCM	GLCM	E-3A	95	95	95	95	95	95	95	95	95	95	95	95
GLCM	GLCM	E-3A	96	96	96	96	96	96	96	96	96	96	96	96
SKAM	SKAM	E-3A	87	90	90	90	90	90	90	90	90	90	90	90
AUDCF	AUDCF	E-3A	89	91	91	91	91	91	91	91	91	91	91	91
UTILITIES	UTILITIES	E-3A	90	91	91	91	91	91	91	91	91	91	91	91
ASTY/TFOCP	ASTY/TFOCP	E-3A	91	91	91	91	91	91	91	91	91	91	91	91
STKTS	STKTS	E-3A	-25	10	-25	25	15	30	25	-10	-20	99	-25	-10
CAPMS	CAPMS	E-3A	45	40	45	45	45	45	45	45	45	45	45	45

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Table D-4. Systems Evaluation Raw Data  
Part 2: SOFTWARE SUPPORT FACILITY (AT DELIVERY)

Scale: - 5 (Low) to + 5 (High) | - 99 (Missing) | + 99 (N/A)

ID SITE	SYSTEM	Evaluation Raw Data											
		SOFTWARE			PER			PER			FAC		
		MAN	TEC	SUP	CON	PER	PER	SYS	SYS	SYS	OTB	OPE	---
93 LANGLEY	TIF1	DC/SR	40	40	30	-30	40	20	30	40	10	30	99
94 LANGLEY	TIF1	II/MARRES/TEREC	10	10	-10	-10	25	-30	10	10	30	-30	-30
95 LANGLEY	407L	HUGHES UTIL	5	5	10	10	10	-10	-10	-5	99	99	10
96 LANGLEY	407L	TERM UTIL	10	10	10	1	99	20	20	99	99	99	-20
97 LANGLEY	407L	FORP/IMPP	10	10	20	5	10	-20	-20	10	10	10	10

Table D-4. Systems Evaluation Raw Data

Part 3: LIFE CYCLE SOFTWARE SUPPORT MANAGEMENT (AT DELIVERY)

		Scale = 5 (Low) to + 5 (High) : - 99 (Missing) : + 99 (N/A)	
ID SITE	SYSTEM	SOFTWARE SYSTEM	AM CON IDE STA CON AUD
		CSS	-40 20 -30 -40 40 -20 -20 -30 -40
1 NOKAD		MEBU	-20 20 20 -10 -10 10 -10 -10 -10
2 NOKAD	NCS	NLS	10 10 10 10 10 10 10 10 10
3 NOKAD	SSC		10 10 10 10 5 10 10 10 5
4 NOKAD	AIR 46	AIR 46	20 20 20 20 15 20 20 20 15
5 WF ALC	ALR 69	ALR 69	-10 5 10 20 -5 10 15 20 15
6 WF ALC	AN/AOL 151	BTG	-10 -10 -20 -30 -40 -30 -30 -40 -40
7 WF ALC	AN/AOL 151	OFF	-10 -10 -20 -30 -40 -30 -30 -40 -40
8 WF ALC	AN/AOL 151	UIT	-40 -10 -40 -50 -30 -30 -40 -10 -40
9 WF ALC	AFK 78	AFK 78	10 20 -99 5 -1 -1 5 5 -10
10 WF ALC	B-52 EVS ATE	HSO-151	10 30 -99 5 5 25 20 -99 20
11 WF ALC	E-7A AVIONICS ATE	AN/OSM-225 (B)	-20 10 -25 30 30 20 20 40 40
12 WF ALC	E-7A AVIONICS ATE	AN/OSM-225 (W)	25 30 25 20 30 20 20 40 40
13 WF ALC	F-15	CC	25 30 20 30 30 20 20 30 40
14 WF ALC	F-15	HADAR	0 0 0 0 0 0 0 0 0
15 WF ALC	F-15	ADS, AIS	-25 10 -20 10 10 20 20 5 20
16 WF ALC	F-15	ASU/OCP	20 15 -99 20 10 40 30 15 25
17 WF ALC	JTIDS	E-3A ANMCS/OCP	25 20 15 29 10 40 30 25 25
18 WF ALC	JTIDS	SP/USER	5 5 -99 -99 40 40 40 40 40
19 WF ALC	JTIDS	SYS EXERCISER	-40 -40 -40 -40 15 -99 0 10 20
20 WF ALC	JTIDS	A1SF	-10 -10 -10 -10 35 30 30 40 40
21 WF ALC	FAVE TACI	DP	42 49 39 42 40 39 38 35 40
22 WF ALC	FAVE TACI	WNC	30 20 15 40 20 30 30 30 30
23 SM ALC	F-111D	WNC	50 20 15 40 20 30 30 30 30
24 SM ALC	F-111F	WNC	50 20 15 40 20 30 30 30 30
25 SM ALC	FB-111A	CFT	-10 5 -10 -10 5 5 5 5 5
26 CASTLE AF8 B-52		WST	5 5 -5 10 -5 5 10 -10 -10
27 CASTLE AF8 B-52		WST	5 5 -5 10 -5 5 10 -10 -10
28 CASTLE AF8 FC-125		WST	-20 -20 -20 5 5 5 5 5 5
29 CASTLE AF8 F-4 TRAINER		F-4 SIMULATOR	-20 -20 -20 5 5 5 5 5 5
30 00 ALC	F-16	FCC	20 20 20 10 10 10 10 10 10
31 00 ALC	F-16	HUD	-10 -20 -10 20 10 20 10 20 10
32 00 ALC	F-16	OFT	30 25 30 25 30 25 30 25 30
33 00 ALC	F-16	FCR	-20 30 30 -50 -40 -20 30 30 20
34 00 ALC	F-16	SMS	2 20 0 -10 0 -6 10 -20 10
35 00 ALC	F-4	MOTS	10 50 10 20 20 10 20 20 20
36 00 ALC	F-4	MOTS	40 40 20 20 10 10 10 10 10
37 00 ALC	F-4E	AN/ARN 101	4 10 1 20 40 40 40 40 40
38 00 ALC	F-4E	AN/ARN 101	40 40 40 45 40 40 40 40 40
39 00 ALC	F-4E	AN/ARN 101	-25 20 20 -39 -25 5 10 10 5
40 00 ALC	F-4E	AN/ARN 101	35 25 40 40 35 40 35 40 40
41 00 ALC	F-4E	AN/ARN 101	20 10 20 -20 10 -20 10 10 10
42 00 ALC	F-4E	AN/ARN 101	10 10 -10 10 10 10 10 10 10
43 00 ALC	F-4E	AN/ARN 101	35 30 30 30 30 30 30 30 30
44 00 ALC	F-4G	AN/ARN 101	20 20 20 20 20 20 20 20 20
45 00 ALC	F-4G	AN/ARN 101	20 25 25 25 25 25 25 25 25
46 00 ALC	F-4G	AN/ARN 101	15 15 15 15 15 15 15 15 15

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Table D-4. Systems Evaluation Raw Data  
Scale: -5 (Low) to +5 (High) + 99 (Missing) + 99 (N/A)

ID SITE	SYSTEM	SOFTWARE SYSTEM	AM CON	AM CON	AM CON	AM CON	AM AUD	AM CON	AM CON	AM PLA	AM O&G	AM DES	AM COD	AM TES	AM INT	AM MAI	AM MAI	AM MAI	A RISK	
47 00-ALC	F-46	AN/ARN-101	35	35	35	35	35	35	35	20	10	-20	-20	-20	-20	-20	20	20	30	0.30
48 00-ALC	F-46	LRF-1/ACM	30	30	30	30	30	30	30	10	20	20	-10	-10	-10	-10	20	20	25	0.40
49 00-ALC	F-46	LRF-1/ACM	30	30	30	30	30	30	30	15	20	15	10	10	10	10	15	20	20	0.50
50 00-ALC	F-46	LRF-1/ACM	20	15	25	20	20	15	20	15	10	10	10	10	10	10	15	15	15	0.70
51 00-ALC	MINUJEMAN	WING VI/HIS-29	5	5	5	5	5	5	5	10	10	10	10	10	10	10	10	10	5	0.75
52 00-ALC	MINUJEMAN	WING VI/HIS-29	99	99	99	99	99	99	99	40	40	40	40	40	40	40	35	40	35	0.10
53 00-ALC	MINUJEMAN	WINGS/HIS-28	10	10	10	10	10	10	25	25	25	25	25	25	25	25	25	25	25	0.60
54 00-ALC	SSAS/CAFS	SSAS/CAFS	50	50	50	50	50	50	50	40	40	40	50	50	50	50	40	50	50	0.20
55 00-ALC	MINUJEMAN	WING V/HEG/RATS	40	40	40	40	40	40	40	30	30	30	30	30	30	30	30	30	40	0.30
56 00-ALC	MINUJEMAN	WING VI/HEG/RATS	40	40	40	40	40	40	40	30	30	30	20	20	20	20	30	30	35	0.30
57 00-ALC	RF-4C	AN/ARN-101	4	10	1	5	1	20	40	40	20	15	10	10	10	10	5	12	40	0.10
58 00-ALC	RF-4C	AN/ARN-101	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	40	0.14
59 00-ALC	RF-4C	AN/ARN-101	-25	-20	-20	-20	-25	-20	5	10	10	5	-10	-5	-5	-5	-5	-5	5	0.60
60 00-ALC	RF-4C	AN/ARN-101	35	25	40	40	40	40	35	40	35	30	30	30	30	30	30	35	35	0.40
61 00-ALC	RF-4C	AN/ARN-101	-20	10	-20	-20	-20	-20	-10	-10	-20	-10	-10	-10	-10	-10	-20	-20	-20	0.90
62 00-ALC	RF-4C	AN/ARN-101	10	10	-10	-10	-10	-10	-10	-10	-10	-10	-10	-10	-10	-10	-20	-20	-20	0.90
63 00-ALC	ALCM	L11	30	20	30	40	20	20	20	20	20	20	20	20	20	20	20	25	-50	0.90
64 00-ALC	ALCM	LPI	40	40	35	40	40	35	40	35	40	35	40	35	40	35	35	35	35	0.90
65 00-ALC	ALCM	OFP	5	-15	5	15	5	15	5	30	30	40	30	30	30	30	30	30	30	0.90
66 00-ALC	ALCM	OFP	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	-50	0.90
67 00-ALC	B-1B	CADC	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	1.00
68 00-ALC	B-1B	CITS	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	1.00
69 00-ALC	B-1B	EMUX	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	1.00
70 00-ALC	B-1B	F/CARS	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	0.60
71 00-ALC	B-1B	INS	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	0.90
72 00-ALC	B-1B	OKS	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99	0.90
73 00-ALC	B-52	BNST	25	40	5	35	40	40	40	30	40	40	40	40	40	40	40	40	40	0.05
74 00-ALC	B-52	E155	10	10	10	15	5	20	20	10	20	15	20	15	30	40	15	10	10	0.10
75 00-ALC	B-52	MC-1 EXEC	20	20	15	20	-99	25	15	20	15	20	15	20	15	10	25	30	0.80	
76 00-ALC	B-52	MC-2 EXEC	-25	10	-25	-10	-10	-30	10	-25	-10	-25	-10	-25	-10	-20	-20	-20	-20	0.75
77 00-ALC	E-7A	INS	99	99	99	99	99	99	99	99	99	99	99	99	99	99	99	99	99	0.95
78 00-ALC	E-7A	OMEGA	10	10	10	10	10	10	5	5	5	5	5	5	5	5	5	5	10	0.80
79 00-ALC	E-7A	SPCP	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	-40	0.95
80 00-ALC	E-7A	SRCF	10	10	10	10	10	10	-5	-5	-5	-5	-5	-5	-5	-5	-5	-5	-5	0.70
81 00-ALC	E-7A	SRSURF	-10	25	-25	-10	15	20	10	25	25	25	25	25	25	25	25	25	25	0.50
82 00-ALC	E-7A	DF-5	-20	10	-40	-30	-30	-30	-10	-30	-20	-20	-20	-20	-20	-20	-20	-20	-20	0.50
83 00-ALC	E-7A	M-010	20	30	20	20	20	20	30	20	30	20	30	20	30	20	20	20	20	0.50
84 00-ALC	E-7A	MP-1	-20	10	-40	-30	-30	-30	-10	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	0.90
85 00-ALC	E-7A	UF-1	-2	5	2	5	2	5	2	-10	-10	-25	-15	-10	-2	-10	2	-10	2	0.50
86 00-ALC	E-7A	WCS	20	10	-40	-30	-30	-30	-10	-30	-30	-30	-30	-30	-30	-30	-30	-30	-30	0.40
87 00-ALC	E-7A	SKAM	25	25	25	25	25	25	25	25	25	25	25	25	25	25	25	25	25	0.40
88 TIM-TR AFB	E-7A	OFF	-20	10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	0.50
89 TIM-TR AFB	E-7A	OFF	-20	10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	0.50
90 LANGLEY JTFUS	E-7A	OFF	-20	10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	0.50
91 LANGLEY SIRIS	E-7A	OFF	-20	10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	0.50
92 LANGLEY TAIS	E-7A	OFF	-20	10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	-10	-50	0.50

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		Table D-4. Systems Evaluation Raw Data											
		Part 3: LIFE CYCLE SOFTWARE SUPPORT MANAGEMENT (AT DELIVERY)											
		Scale: -50 (Low) to +50 (High) + 99 (N/A)											
ID SITE	SYSTEM	AM CON	AM IDE	AM STA	AM CON								
93 LANGLEY	TIP1	-30	-50	10	-40	10	-40	30	-10	30	-10	30	-10
94 LANGLEY	TIP1	-10	10	15	10	-10	15	10	10	-10	10	-10	0
95 LANGLEY	407L	10	10	10	10	5	5	5	1	10	5	10	-40
96 LANGLEY	4071	10	10	10	10	1	10	10	5	20	5	10	50
97 LANGLEY	407L	15	25	20	20	20	20	25	15	10	20	20	20

Table D-4. Systems Evaluation Raw Data  
Part 4: SOFTWARE PRODUCT MAINTAINABILITY (CURRENT)

ID SITE	SYSTEM	Scale: - 50 (low) to + 50 (high) + 99 (Missing) + 99 (N/A)										
		AP DOC MOD	AP DOC DES	AP DOC CON	AP DOC SIM	AP EXP	AP INS	AP SRC MOD	AP SRC DES	AP SRC CON	AP SRC SIM	AP SRC EXP
1 NORAD	CSS	-20	10	-5	-10	10	-15	-5	-8	-20	15	-5
2 NORAD	MEBU	-20	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
3 NORAD	NCS	-10	-99	-99	-99	-99	-99	-99	-99	-99	-99	-99
4 NORAD	SSC	-20	25	25	25	25	25	5	10	15	5	10
5 WR-ALC	ALR-46	-20	25	30	20	24	15	29	10	10	18	5
6 WR-ALC	ALR-69	-20	5	30	29	30	40	35	25	20	10	25
7 WR-ALC	BIG	-40	-40	-20	-30	39	-50	30	-10	-40	-10	-25
8 WR-ALC	OFP	15	-30	-10	39	20	40	10	-30	10	20	30
9 WR-ALC	UNIT	10	-20	20	-20	-20	-40	-50	-10	-10	-20	-20
10 WR-ALC	AFK-78	70	30	25	50	0	50	25	15	20	50	0
11 WR-ALC	ASD 151	40	45	25	40	35	35	20	5	20	1	20
12 WR-ALC	E-3A AVIONICS ATE	10	-20	10	10	10	10	10	-10	10	10	10
13 WR-ALC	E-3A AVIONICS ATE	30	30	10	20	25	40	10	30	40	20	10
14 WR-ALC	F-15	25	20	20	40	10	20	30	30	30	35	40
15 WR-ALC	RADAR	40	40	35	45	35	50	40	45	50	40	40
16 WR-ALC	F-15 AVIONICS ATE	10	20	20	40	30	10	-10	5	20	5	15
17 WR-ALC	JTIDS	35	39	39	39	39	35	30	30	30	30	30
18 WR-ALC	JTIDS	35	30	30	30	30	30	35	30	30	30	30
19 WR-ALC	JTIDS	15	30	0	30	-10	10	10	0	30	10	-10
20 WR-ALC	SYS EXERCISER	-15	-20	-10	-10	0	-99	-15	-20	-10	-10	-99
21 WR-ALC	AISF	45	35	35	40	40	35	45	45	40	45	50
22 WR-ALC	OFF	35	45	40	35	40	40	38	42	40	38	30
23 SM ALC	WNC	40	25	25	30	30	40	-99	25	30	10	-20
24 SM ALC	WNC	40	25	25	30	30	40	-99	25	30	10	-20
25 SM ALC	FB-111A	40	25	25	30	30	40	-99	25	30	10	-20
26 CASTLE AFB B-52	CPT	15	20	5	10	20	-50	10	20	-5	-5	20
27 CASTLE AFB B-52	WST	10	20	5	-20	-10	0	10	10	-20	5	10
28 CASTLE AFB B-115	T-4 SIMULATOR	10	20	5	-20	-10	0	10	10	-20	5	10
29 CASTLE AFB T-4 TRAINER	F-16	10	25	10	25	10	-5	-50	5	-10	5	-50
30 OO-ALC	HUD	15	30	10	5	10	10	5	30	25	10	20
31 OO-ALC	F-16	10	40	-20	15	5	45	20	15	40	-15	5
32 OO-ALC	F-16	-30	-10	-40	-30	10	-20	-35	-10	-40	-30	-40
33 OO-ALC	F-16	5	20	20	20	20	20	20	10	20	20	20
34 OO-ALC	F-4	25	35	25	30	19	25	10	30	35	20	20
35 OO-ALC	F-4	10	30	10	30	20	30	10	40	10	-10	20
36 OO-ALC	F-4E	20	45	10	20	20	25	10	10	20	10	-10
37 OO-ALC	AN/ARN-101	30	50	30	25	25	30	-40	10	25	20	-10
38 OO-ALC	F-4E	-99	10	20	-10	35	-40	25	10	20	-10	20
39 OO-ALC	F-4E	5	20	20	20	20	20	20	10	20	20	20
40 OO-ALC	F-4	5	40	5	35	5	20	15	20	45	30	5
41 OO-ALC	F-4E	-10	30	-10	30	-10	20	-20	20	30	10	-20
42 OO-ALC	F-4E	-10	10	-20	10	10	10	10	10	10	10	10
43 OO-ALC	F-46	30	35	40	35	-10	35	40	40	40	40	40
44 OO-ALC	F-46	30	35	30	10	15	30	35	35	35	10	10
45 OO-ALC	F-46	28	35	38	40	40	45	30	35	40	45	30
46 OO-ALC	F-46	25	30	25	40	35	-5	-5	30	30	40	5

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Table D-4. Systems Evaluation Raw Data

Part 4: SOFTWARE PRODUCT MAINTAINABILITY (CURRENT)

ID SITE	SYSTEM	Scales - 5 (Low) to + 50 (High) - 99 (Missing) + 99 (N/A)									
		AP DOC	AP DOC	AP DOC	AP SRC						
		MOD	DES	CON	SIM	EXP	INS	MOD	DES	CON	INS
47 00-ALC	F-46										
48 00-ALC	F-46										
49 00-ALC	F-46										
50 00-ALC	F-46										
51 00-ALC	MINUTEMAN										
52 00-ALC	MINUTEMAN										
53 00-ALC	MINUTEMAN										
54 00-ALC	MINUTEMAN										
55 00-ALC	MINUTEMAN										
56 00-ALC	MINUTEMAN										
57 00-ALC	RF-4C										
58 00-ALC	RF-4C										
59 00-ALC	RF-4C										
60 00-ALC	RF-4C										
61 00-ALC	RF-4C										
62 00-ALC	RF-4C										
63 00-ALC	ALCM										
64 00-ALC	ALCM										
65 00-ALC	ALCM										
66 00-ALC	ALCM										
67 00-ALC	B-1B										
68 00-ALC	B-1B										
69 00-ALC	B-1B										
70 00-ALC	B-1B										
71 00-ALC	B-1B										
72 00-ALC	B-1B										
73 00-ALC	B-52										
74 00-ALC	B-52										
75 00-ALC	B-52										
76 00-ALC	B-52										
77 00-ALC	E-1A										
78 00-ALC	E-1A										
79 00-ALC	E-1A										
80 00-ALC	E-1A										
81 00-ALC	E-1A										
82 00-ALC	GCM										
83 00-ALC	GCM										
84 00-ALC	GCM										
85 00-ALC	GCM										
86 00-ALC	GCM										
87 00-ALC	GCM										
88 100-AR	MINUTEMAN										
89 100-AR	MINUTEMAN										
90 100-AR	MINUTEMAN										
91 100-AR	SIRIS										
92 100-AR	SIRIS										

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Part I: SOFTWARE PRODUCT MAINTAINABILITY (CURRENT)

ID SITE	SYSTEM	Scale: - 50 (Low) to + 50 (High) : - 99 (Missing) : + 99 (N/A)									
		AF	AF	AF	AF	AP	AP	AP	AP	AP	AP
	SOFTWARE SYSTEM	DOC MOD	DOC DES	DOC CON	SIM EXP	SIM INS	MOD DEF	SRC CON	SRC SIM	SRC EXP	A
93 LANGLEY	TIF1	20	20	30	10	10	25	25	40	10	20
94 LANGLEY	TIF1	10	30	-5	30	15	-20	40	15	25	30
95 LANGLEY	407L	5	10	10	5	5	0	-30	-10	-5	-50
96 LANGLEY	407L	30	40	40	40	30	30	40	40	40	-40
97 LANGLEY	407L	40	45	30	40	25	40	45	30	30	40
	DC/SR										
	II/MARFES/TEREC										
	HUGHES UTIL										
	IBM UTIL										
	10KF/1MF-P										

Table D-4. Systems Evaluation Raw Data  
Part 5: SOFTWARE SUPPORT FACILITY (CURRENT)

ID SITE	SYSTEM	Scales - 5(Low) to + 5(High) + 99(Missing) + 99(N/A)									
		AE	AE	AE	AE	AE	AE	AE	AE	AE	AE
	SOFTWARE SYSTEM	PER MAN	PER TEC	PER SUP	PER CON	PER SYB	PER BEN	PER LAB	SYB OTH	FAC OFF	
1 NOKAD	CSS	20	20	10	-10	-20	-99	-99	-99	-30	-30
2 NOKAD	MESU	-99	-99	10	10	-20	-20	-20	-99	-99	-99
3 NOKAD	NCS	15	15	15	15	15	25	99	99	25	25
4 NOKAD	SSC	15	15	15	15	15	25	99	99	25	25
5 WR AL C	ALR-46	20	10	20	15	99	35	40	40	20	10
6 WR AL C	ALR-69	29	40	35	25	99	45	40	45	99	45
7 WR AL C	B1B	-16	10	-20	10	10	20	30	40	40	40
8 WR AL C	OFF	29	10	10	-20	10	10	30	20	40	40
9 WR AL C	UUT	-20	10	-50	-40	-10	-15	-10	-10	-10	-10
10 WR AL C	ARF-38	20	25	20	25	99	20	0	25	25	25
11 WR AL C	ASQ-131	35	40	40	40	40	20	20	30	30	30
12 WR AL C	E TA AVIATICS ATE	40	40	40	40	40	25	20	5	45	20
13 WR AL C	E TA AVIATICS ATE	40	40	40	40	40	25	20	5	45	20
14 WR AL C	CC-10	20	10	20	10	10	40	30	40	30	30
15 WR AL C	RADAR	10	0	10	35	45	25	35	35	40	40
16 WR AL C	ADS, A16	25	20	10	20	99	25	10	20	5	10
17 WR AL C	ASLT/OCF	25	25	20	15	15	10	5	25	20	0
18 WR AL C	E TA AWACS/OCF	25	20	20	15	15	10	5	25	20	0
19 WR AL C	SP/USER	35	-94	15	-99	99	-5	-5	20	-10	30
20 WR AL C	SYS EXERCISE	25	99	25	99	99	-20	99	99	99	99
21 WR AL C	AISF	20	40	20	25	40	40	45	45	25	40
22 WR AL C	OFF/TAC	25	31	30	25	40	40	38	38	35	35
23 SM AL C	WNC	40	20	40	30	20	20	20	20	30	30
24 SM AL C	WNC	40	20	40	30	30	20	20	20	20	20
25 SM AL C	WNC	40	20	40	30	30	20	20	20	20	20
26 CASTLE AFB B-52	CFT	20	30	30	-30	5	5	-99	-99	-99	-99
27 CASTLE AFB B-52	WST	10	20	10	10	10	10	20	10	5	10
28 CASTLE AFB B-52	WST	15	20	20	10	10	10	20	10	5	10
29 CASTLE AFB T-4 TRAINER	I-4 SIMULATOR	25	30	30	25	20	20	20	20	20	20
30 JG-AL C	F-10	0	0	-10	0	20	30	30	30	-30	0
31 OO-AL C	F-16	-15	-20	-20	-20	25	30	30	30	-30	-10
32 OO-AL C	F-16	30	-20	40	20	99	45	30	50	45	45
33 OO-AL C	F-16	-40	0	-30	-40	-50	-30	50	-30	-30	-30
34 OO-AL C	F-16	-5	10	-10	10	-20	8	-10	0	-20	0
35 OO-AL C	F-4	25	40	45	25	30	30	30	20	-25	-25
36 OO-AL C	F-4	45	45	40	40	40	30	10	20	20	40
37 OO-AL C	F-4E	28	40	15	10	45	25	99	15	-99	40
38 OO-AL C	F-4E	40	40	40	40	40	45	45	45	45	45
39 OO-AL C	F-4E	78	40	40	40	40	40	45	45	45	45
40 OO-AL C	F-4E	35	40	40	40	40	25	10	5	10	10
41 OO-AL C	F-4E	40	40	45	40	40	20	5	5	5	5
42 OO-AL C	F-4E	25	75	35	20	30	30	20	20	20	20
43 OO-AL C	F-4E	20	30	30	30	30	10	10	10	10	10
44 OO-AL C	F-4E	25	-99	-5	20	-25	-10	-10	-10	-10	-10
45 OO-AL C	F-4E	24	25	5	30	30	-5	-5	-5	-5	-5
46 OO-AL C	F-4E	21	5	5	30	30	25	25	25	25	25
47 OO-AL C	F-4E	21	5	5	30	30	20	20	20	20	20

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Table D-4. System Evaluation Raw Data  
Part 5: SOFTWARE SUPPORT FACILITY (CURRENT)

ID SITE	SYSTEM	Scale = 50 (Low) to + 50 (High) : - 99 (Missing) : + 99 (N/A)									
		AE PER MAN	AE PER TEC	AE PER SUF	AE PER CON	AE SYS HOS	AE SYS BEN	AE SYG LAB	AE SYS OTH	AE FAC DPE	AE FAC OFF
47	00-ALC	F-4G	AN/ARN-101	-20	-20	-5	-20	-10	-10	-99	-99
48	00-ALC	F-4G	LRU-1/ACM	30	20	40	20	99	30	20	30
49	00-ALC	F-4G	LRU-1/ACM	45	40	45	45	45	45	40	45
50	00-ALC	F-4G	LRU-1/ACM	25	25	25	25	25	20	15	30
51	00-ALC	MINUTEMAN	WING LI/2015	38	25	40	45	40	40	45	45
52	00-ALC	MINUTEMAN	WING VI/HS-29	45	25	45	45	40	35	45	45
53	00-ALC	MINUTEMAN	WINGS/HS-28	38	25	40	45	40	40	45	45
54	00-ALC	MINUTEMAN	SSAS/CAPS	50	30	20	50	20	10	99	30
55	00-ALC	MINUTEMAN	WING V/HEG/KATS	20	20	20	30	10	20	10	40
56	00-ALC	MINUTEMAN	WING VI/HEG/RATS	30	20	-99	20	30	40	-70	40
57	00-ALC	RF-4C	AN/ARN-101	28	40	15	10	45	25	25	15
58	00-ALC	RF-4C	AN/ARN-101	40	40	40	40	45	20	45	40
59	00-ALC	RF-4C	AN/ARN-101	35	40	40	25	40	10	25	10
60	00-ALC	RF-4C	AN/ARN-101	40	40	45	40	40	10	25	10
61	00-ALC	RF-4C	AN/ARN-101	25	35	20	30	30	-20	99	20
62	00-ALC	RF-4C	LIT	30	20	30	30	40	-10	-20	99
63	00-ALC	ALCM	ALCM	40	40	45	35	50	50	40	40
64	00-ALC	ALCM	LFT	35	35	35	35	40	40	40	40
65	00-ALC	ALCM	OFF	25	20	30	0	15	20	20	20
66	00-ALC	ALCM	OFF	-15	-50	25	50	25	10	5	5
67	00-ALC	B-1B	CADC	22	30	10	15	-50	-99	-99	-99
68	00-ALC	B-1B	CINX	22	30	20	15	-20	99	10	-30
69	00-ALC	B-1B	EMUX	22	30	10	15	-20	99	-99	-30
70	00-ALC	B-1B	F/CGMS	22	30	20	15	-20	99	-99	-30
71	00-ALC	B-1B	INS	22	31	-20	10	-99	-99	-99	-30
72	00-ALC	B-1B	ORS	25	30	-10	10	-50	-99	-99	-30
73	00-ALC	B-1B	BNST	35	40	35	50	10	30	45	5
74	00-ALC	B-52	FTSS	75	40	30	35	40	47	50	50
75	00-ALC	B-52	MC-1 EXEC	20	25	30	25	40	30	40	40
76	00-ALC	B-52	MC-2 EXEC	35	-99	35	10	99	20	35	35
77	00-ALC	E-3A	INS	20	10	20	30	-10	30	-50	-10
78	00-ALC	E-3A	OMEGA	30	20	40	40	30	25	30	-40
79	00-ALC	E-3A	SMCP	-20	-30	5	-20	10	10	10	-10
80	00-ALC	E-3A	SRCP	5	-10	30	-5	0	10	-10	-5
81	00-ALC	E-3A	SRGSCP	30	30	25	-5	25	25	99	35
82	00-ALC	GLCM	D/S	-35	-40	-40	-30	20	10	5	45
83	00-ALC	GLCM	K-DIO	20	10	30	30	30	20	40	10
84	00-ALC	GLCM	MFI	10	-20	10	-10	15	10	10	-5
85	00-ALC	GLCM	OFF	30	25	30	30	99	30	25	10
86	00-ALC	GLCM	WCS	-20	-20	-5	-20	5	10	20	10
87	00-ALC	SKRM	OFF	-30	15	-10	-25	25	10	5	10
88	110 ER AFB	E-3A	ANC/P	30	20	50	20	99	20	40	10
89	110 ER AFB	E-3A	UTILITIES	40	30	40	40	99	50	99	-10
90	110 ER AFB	E-3A	ASIT/FOCF	30	15	35	30	5	20	20	10
91	110 ER AFB	E-3A	SIRFS	-25	-10	-25	-25	10	25	25	10
92	110 ER AFB	E-3A	TACFS	40	40	40	40	99	40	50	10

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Table D-4. Systems Evaluation Raw Data  
Part 5: SOFTWARE SUPPORT FACILITY (CURRENT)

		Scale: -5 (Low) to +50 (High) + 99 (Missing) + 99 (N/A)															
ID SITE	SYSTEM	SOFTWARE SYSTEM	AE PER MAN	AE PER TEC	AE PER SURF	AE PER CON	AE SYS HOS	AE SYS BEN	AE SYS LAB	AE SYS OTH	AE SYS OPE	AE SYS OTH	AE FAC	AE FAC	AE OFF	AE ENV	AE IRDN
93 LANGLEY	TIP1	DC/SR	10	20	-10	10	10	20	10	99	30	20	99	20	10	30	20
94 LANGLEY	TIP1	11/MARRES/TEREC	25	25	10	10	35	20	20	10	0	0	10	20	20	20	20
95 LANGLEY	407L	HUGHES UTIL	5	5	10	10	10	-10	-10	-5	99	99	99	10	10	10	5
96 LANGLEY	407L	IBM UTIL	10	10	10	1	99	20	20	99	99	99	-20	-20	-20	-5	
97 LANGLEY	407L	10RP/IMPF	30	35	30	50	35	40	10	35	40	99	40	40	35	45	40

Table D-4. Systems Evaluation Raw Data  
Part 6: LIFT CYCLE SOFTWARE SUPPORT MANAGEMENT (CURRENT)

ID SITE		SYSTEM		SOFTWARE SYSTEM		Scale: -50 (Low) to +50 (High)		1 - 99 (Missing)		1 + 99 (N/A)	
						AM CON IDE STA		AM CON AUD		AM CON STA	
1	NORAD	CSS		15	-30	15	-40	20	20	25	10
2	NORAD	MEBU		30	30	30	-10	-10	-10	-10	-10
3	NORAD	NCS		20	20	20	-10	-10	-10	-10	-10
4	NORAD	SSC		25	25	25	20	30	20	20	-99
5	WR-ALC	ALK-46		20	20	20	20	28	30	35	38
6	WR-ALC	ALK-69		40	40	30	45	30	35	30	30
7	WR ALC	BTG		-30	-20	30	10	50	10	20	40
8	WR ALC	OFF		-30	-20	30	10	50	10	20	40
9	WR ALC	UUT		-30	-20	30	10	50	10	20	40
10	WR-ALC	AFR-38		15	20	-99	20	8	15	10	10
11	WR ALC	H-52 EVS ATE		49	40	-99	45	45	50	25	25
12	WR ALC	E-3A AVIONICS ATE		-20	-10	10	-25	-30	20	40	40
13	WR ALC	E-3A AVIONICS ATE		-25	-30	25	20	30	20	40	40
14	WR ALC	F-15		25	30	30	20	20	20	40	40
15	WR ALC	KADAK		30	30	30	20	20	20	30	30
16	WR ALC	ADS-AIS		20	5	30	30	40	30	30	30
17	WR ALC	JTIDS		20	15	-99	20	10	40	30	30
18	WR ALC	JTIDS		25	20	15	20	10	40	30	30
19	WR ALC	JTIDS		5	-99	-99	40	40	40	30	30
20	WR ALC	SYS EXERCISER		-40	-40	-40	15	-99	10	20	0
21	WR-ALC	FAVE FAC		45	45	45	35	30	40	40	45
22	WR-ALC	FAVE TACI		40	42	40	43	40	38	39	40
23	SM-ALC	F-111D		20	20	10	30	20	20	25	25
24	SM ALC	F-111F		20	10	30	20	20	20	10	25
25	SM-ALC	FB-111A		20	10	30	20	20	20	10	25
26	CASTLE AFB B-52	CFT		10	10	10	15	15	15	15	15
27	CASTLE AFB B-52	WST		15	10	5	10	15	5	10	10
28	CASTLE AFB I-C-125	WST		15	10	5	10	15	5	10	10
29	CASTLE AFB T-4 TRAINER	T-4 SIMULATOR		20	20	20	15	15	15	15	15
30	OO-ALC	F-16		20	20	20	20	20	20	20	20
31	OO-ALC	HUD		-10	-10	-10	20	10	20	10	-10
32	OO-ALC	OFT		30	25	30	40	40	25	35	30
33	OO-ALC	FCR		-20	30	-50	-40	-20	-20	-30	-20
34	OO-ALC	SMS		18	30	10	15	15	20	20	16
35	OO-ALC	MOTS		30	50	25	40	35	40	30	30
36	OO-ALC	MDTS		40	30	40	40	40	40	30	30
37	OO ALC	AN/ARN-101		4	10	1	5	1	20	40	40
38	OO ALC	F-4E		40	40	40	45	40	30	35	40
39	OO ALC	F-4E		20	20	15	25	20	25	5	15
40	OO ALC	F 4E		35	25	40	40	35	50	30	30
41	OO ALC	F 4E		30	20	20	25	20	20	20	20
42	OO ALC	F-4E		20	30	30	30	30	30	30	30
43	OO ALC	F 4G		35	35	35	30	25	30	35	30
44	OO ALC	F 4G		15	25	25	30	25	20	20	25
45	OO ALC	F 4G		20	25	25	30	25	20	25	25
46	OO ALC	F 4G		40	40	40	40	40	40	40	40

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 Table 0-4. Systems Evaluation Raw Data  
 Partial LIFE CYCLE SOFTWARE SUPPORT MANAGEMENT (CURRENT)

ID SITE	SYSTEM	Software System			Scale: - Software to + Software			Scale: - 99(Missing) to + 99(N/A)			A			A			A		
		AM CON	AM IDE	AM STA	AM CON	AM AUD	AM PLA	AM MAI	MAI DES	MAI COD	MAI TES	MAI INT	MAN AGE	MAN PORT	SUFT RISK				
47 00-ALC	F-4G	40	40	40	25	20	20	20	30	30	30	-5	-5	-5	30	30	30	30	30
48 00-ALC	F-4G	30	30	30	25	20	20	20	35	35	40	-5	-5	-5	20	20	20	20	20
49 00-ALC	F-4G	35	40	35	35	35	35	30	30	35	35	-5	-5	-5	35	35	35	35	35
50 00-ALC	F-4G	20	15	25	20	25	20	25	10	10	10	-5	-5	-5	25	25	25	25	25
51 00-ALC	MINUTEMAN	5	5	5	5	5	5	15	15	15	15	15	15	15	45	45	45	45	45
52 00-ALC	MINUTEMAN	WING VI/HS-24	40	40	40	40	40	40	40	40	40	40	40	40	45	45	45	45	45
53 00-ALC	MINUTEMAN	WINGS/HS-28	10	10	10	10	10	10	25	25	25	25	25	25	25	25	25	25	25
54 00-ALC	MINUTEMAN	SSAS/CAFS	20	16	10	30	30	10	30	40	40	40	40	40	45	45	45	45	45
55 00-ALC	MINUTEMAN	WING V/HB/KATS	40	40	40	40	40	40	30	30	30	30	30	30	30	35	35	35	35
56 00-ALC	MINUTEMAN	WING V/HEG/KATS	40	40	40	40	40	40	30	30	30	30	30	30	30	35	35	35	35
57 00-ALC	RF-4C	AN/ARN-101	4	10	1	5	1	20	40	40	40	40	40	40	5	12	40	40	40
58 00-ALC	RF-4C	AN/ARN-101	40	40	40	45	40	40	30	35	35	35	35	35	40	40	40	40	40
59 00-ALC	RF-4C	AN/ARN-101	20	20	15	25	20	15	25	5	5	5	5	5	15	15	15	15	15
60 00-ALC	RF-4C	AN/ARN-101	35	25	40	40	35	40	35	30	30	30	30	30	35	35	35	35	35
61 00-ALC	RF-4C	AN/ARN-101	30	20	20	20	20	20	20	20	20	20	20	20	20	20	20	20	20
62 00-ALC	RF-4C	AN/ARN-101	50	20	30	30	30	30	15	10	20	10	20	10	10	10	10	10	10
63 00-ALC	ALCM	LIT	30	30	40	20	40	30	30	35	40	40	40	40	45	45	45	45	45
64 00-ALC	ALCM	LFI	30	40	35	20	20	30	30	40	40	40	40	40	20	20	20	20	20
65 00-ALC	ALCM	OFFP	30	25	15	35	10	30	30	40	40	40	40	40	25	25	25	25	25
66 00-ALC	ALCM	OPP	10	5	10	15	10	5	-40	-50	15	40	40	35	5	-30	-30	-30	-30
67 00-ALC	B-1B	CADC	24	30	5	-10	17	-15	-10	-15	-10	-10	-10	-10	-30	-5	-10	-10	-10
68 00-ALC	B-1B	CITS	29	30	5	30	17	-30	10	5	10	20	20	20	-40	-40	-40	-40	-40
69 00-ALC	B-1B	EMUX	26	30	5	10	17	-18	10	18	10	18	10	18	-30	-30	-30	-30	-30
70 00-ALC	B-1B	F/GMS	24	30	5	-10	17	-10	10	5	-10	5	-10	5	-10	-15	-15	-15	-15
71 00-ALC	B-1B	INS	28	30	5	20	17	12	10	10	15	20	20	20	-10	-10	-10	-10	-10
72 00-ALC	B-1B	ORS	-20	1	-10	-30	17	-10	10	7	8	28	28	28	-37	-37	-37	-37	-37
73 00-ALC	B-52	BNSF	25	40	5	35	35	40	40	30	40	40	40	40	-45	-45	-45	-45	-45
74 00-ALC	B-52	FTSS	10	10	15	5	20	20	10	20	15	30	40	40	15	15	15	15	15
75 00-ALC	B-52	MC-1 EXEC	40	50	20	35	-99	40	30	35	40	45	45	45	35	35	35	35	35
76 00-ALC	B-52	MC-2 EXEC	-25	14	-25	-10	-10	-10	30	10	-25	-10	-10	-10	-20	-20	-20	-20	-20
77 00-ALC	E-3A	INS	99	99	99	99	99	99	99	99	99	99	99	99	99	99	99	99	99
78 00-ALC	E-3A	OMEGA	20	20	15	20	10	5	5	10	10	5	5	5	5	5	5	5	5
79 00-ALC	E-3A	SMCF	-10	-10	10	-40	5	5	5	5	5	5	5	5	5	5	5	5	5
80 00-ALC	E-3A	SRCF	10	10	10	10	-5	-5	-5	-5	-5	-5	-5	-5	-5	-5	-5	-5	-5
81 00-ALC	E-3A	SKSDF	25	5	15	5	30	20	30	25	25	10	10	10	25	25	25	25	25
82 00-ALC	E-3A	DFS	-20	10	-40	-30	-10	-30	-10	-30	-10	-20	-10	-20	-20	-20	-20	-20	-20
83 00-ALC	E-3A	M DID	-25	10	-30	-40	-12	-5	-40	-12	-5	-40	-10	-10	-10	-18	-18	-18	-18
84 00-ALC	GLCM	MFT	-20	10	-40	-30	-10	-30	-10	-30	-10	-30	-10	-10	-15	-15	-15	-15	-15
85 00-ALC	GLCM	OFF	2	2	5	-5	-10	15	10	20	15	20	15	20	10	10	10	10	10
86 00-ALC	GLCM	WCS	-20	10	-40	-30	-10	-30	-10	-30	-10	-30	-10	-10	-25	-25	-25	-25	-25
87 00-ALC	SKAM	OFF	25	25	25	25	25	25	25	25	25	25	25	25	15	15	15	15	15
88 00-ALC	SKAM	ACOF	20	20	10	10	30	20	30	20	30	20	30	20	-10	-10	-10	-10	-10
89 00-ALC	UTILITIES	20	30	20	20	10	20	20	20	20	20	20	20	20	30	30	30	30	30
90 00-ALC	UTILITIES	ABY/TFCP	5	10	5	-10	-25	15	15	15	15	15	15	15	20	20	20	20	20
91 00-ALC	STRIS	40	25	25	25	25	25	25	25	25	25	25	25	25	30	30	30	30	30
92 00-ALC	CAFMS	40	40	40	40	40	40	40	40	40	40	40	40	40	35	35	35	35	35

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Table D-4. Systems Evaluation Raw Data  
Part b: LIFE CYCLE SOFTWARE SUPPORT MANAGEMENT (CURRENT)

Scales - 50 (Low) to + 50 (High) 1 - 99 (Missing) 1 + 99 (N/A)

ID SITE	SYSTEM	SOFTWARE SYSTEM	AM CON IDE STA	AM CON CON AUD	AM CON CON AUD	AM MAI PLA ORG	AM MAI MAI DES COD	AM MAI TES	AM MAN AGES	AM MAI INT	AM SUP-PORT	A RISK
93 LANGLEY	TIP1	DC/SR	50 40	50 40	50 40	50 45	50 45	50 40	-10 40	30 45	30 35	30 0.75
94 LANGLEY	TIF1	11/MARRES/TEREC	45 10	45 10	40 10	45 10	45 5	45 5	-10 5	30 10	30 5	20 0.50
95 LANGLEY	407L	HUGHES UTIL	10 10	10 10	10 10	10 10	10 5	10 5	-10 10	30 10	30 5	20 0.90
96 LANGLEY	407L	IBM UTIL	10 10	10 10	10 10	10 10	10 5	10 5	-10 5	30 10	30 5	20 0.90
97 LANGLEY	407L	TORF/1MPP	25 25	25 25	20 20	25 25	25 25	30 30	40 40	35 40	40 40	40 0.40

Table D-5. Systems Maintenance Block Release Raw Data

#	SOFTWARE SYSTEM	RLS ID	RLS START DATE	ENGR COMP DATE	% MOS PERS S/W RLS EST	% MOS PERS S/W RLS ACT	TOT CHNG CRRK	% HIGH CONV	% MED CONV	% LOW CONV	NO. ERER	NO. CPLX	NO. CPLX	NO. PRIOR	NO. URG NORM
1	NORAD CSS	CS1	03/07/79	02/07/80	11.00	71	100	55	0	0	75	72	3	17	25
2	NORAD CSb	CS2	08/03/80	10/03/80	2.00	71	100	5	0	0	72	1	1	0	0
3	NOKAD CSb	CS3	03/13/80	02/13/81	11.00	74	100	53	0	0	72	65	7	0	15
4	NOKAD CSb	CS4	05/27/80	04/27/81	11.00	74	100	34	0	0	36	28	8	12	16
5	NOKAD CSb	CS5	09/03/80	08/03/81	11.00	74	100	45	0	0	59	39	20	0	29
6	NOKAD CSb	CS6	12/24/80	11/25/81	11.00	74	100	33	0	0	62	54	8	0	25
7	NOKAD CSb	CS7	04/15/81	03/15/82	11.00	75	100	57	0	0	68	56	12	0	28
8	NOKAD CSb	CS8	08/30/81	07/30/82	11.00	75	100	41	0	0	50	39	11	0	30
9	NOKAD CSb	CS9	01/15/82	12/15/82	11.00	75	100	45	0	0	43	30	13	0	11
10	NOKAD CSb	CS10	06/26/82	05/26/83	11.00	73	100	44	0	0	66	54	12	0	25
11	NOKAD CSb	CS11	05/26/83	07/01/83	11.10	73	100	5	0	0	66	51	12	0	12
12	NOKAD MDS	MDS1	11/14/82	10/14/83	11.00	73	100	42	0	0	65	55	10	0	33
13	NOKAD MDS	MDS2	03/29/83	02/29/84	11.00	73	100	42	0	0	79	61	18	0	43
14	NOKAD MDS	MDS3	09/01/83	08/01/84	11.00	73	100	45	0	0	58	50	8	0	48
15	NOKAD MDS	MDS4	08/01/84	09/01/84	11.00	73	100	5	0	0	1	1	0	0	42
16	NOKAD MDS	MDS5	02/01/84	01/01/85	11.00	73	100	45	0	0	65	51	14	0	0
17	NOKAD MDS	MDS6	06/01/84	05/01/85	11.00	61	100	45	0	0	35	27	8	0	23
18	NOKAD MDS	MDS7	08/23/78	03/23/79	11.00	8	100	45	0	0	4	4	1	1	2
19	NOKAD MDS	MDS8	02/13/80	01/13/81	11.00	10	100	32	0	0	12	9	3	0	10
20	NOKAD MDS	MDS9	01/22/80	10/22/81	11.00	10	100	28	0	0	3	0	0	0	5
21	NOKAD MDS	MDS10	05/23/81	04/23/82	11.00	9	100	5	0	0	2	1	0	0	2
22	NOKAD MDS	MDS11	DCE	11/04/82	10/04/83	11.00	8	100	5	0	0	1	1	0	42
23	NOKAD MDS	MDS12	DGA	01/01/79	12/01/79	11.00	8	100	45	0	0	35	27	8	0
24	NOKAD MDS	MDS13	DGB	11/22/80	10/22/81	11.00	10	100	34	0	0	4	4	0	10
25	NOKAD MDS	MDS14	DGB1	10/22/82	12/14/82	1.80	9	100	5	0	0	12	9	3	0
26	NOKAD MDS	MDS15	I1FA	04/29/78	03/29/79	11.00	8	100	45	0	0	3	0	0	5
27	NOKAD MDS	MDS16	I1FA1	03/29/79	06/01/79	2.00	8	100	5	0	0	2	1	0	2
28	NOKAD MDS	MDS17	IFB	11/21/79	10/21/80	11.00	8	100	5	0	0	1	1	0	10
29	NOKAD MDS	MDS18	IFC	08/01/80	07/01/81	11.00	9	100	34	0	0	16	9	7	4
30	NOKAD MDS	MDS19	IFC1	08/21/80	07/21/81	11.00	10	100	5	0	0	12	3	9	2
31	NOKAD MDS	MDS20	IFD	11/22/80	10/22/81	11.00	10	100	5	0	0	1	1	1	5
32	NOKAD MDS	MDS21	IFD1	10/22/81	01/06/82	2.50	9	100	5	0	0	2	0	0	15
33	NOKAD MDS	MDS22	IFE	05/23/81	04/23/82	11.00	9	100	5	0	0	1	1	0	0
34	NOKAD MDS	MDS23	IFF	12/15/81	11/15/82	11.00	9	100	32	0	0	7	6	1	4
35	NOKAD MDS	MDS24	IFG	11/04/82	10/04/83	11.00	8	100	5	0	0	12	3	4	10
36	NOKAD MEBU	MEBU1	B	12/30/78	11/30/79	11.00	20	100	45	0	0	2	0	0	2
37	NOKAD MEBU	MEBU2	C	05/10/79	04/10/80	11.00	20	100	64	0	0	96	74	22	0
38	NOKAD MEBU	MEBU3	C1	05/29/80	07/29/80	2.00	20	100	100	0	0	26	14	12	0
39	NOKAD MEBU	MEBU4	C1A	02/06/81	03/06/81	1.00	18	100	33	0	0	4	2	0	2
40	NOKAD MEBU	MEBU5	C1E	09/01/80	08/01/81	11.00	18	100	64	0	0	104	81	23	0
41	NOKAD MEBU	MEBU6	E	12/24/80	11/24/81	11.00	18	100	41	0	0	40	34	6	17
42	NOKAD MEBU	MEBU7	F	05/30/81	04/30/82	11.00	17	100	36	0	0	42	36	6	15
43	NOKAD MEBU	MEBU8	G	08/29/81	07/29/82	11.00	17	100	39	0	0	13	10	3	16
44	NOKAD MEBU	MEBU9	H1	01/15/82	12/15/82	11.00	17	100	42	0	0	40	28	12	0
45	NOKAD MEBU	MEBU10	H1A	08/05/82	07/05/83	11.00	17	100	44	0	0	36	24	12	8
46	NOKAD MEBU	MEBU11	J	11/17/82	10/17/83	11.00	17	100	44	0	0	41	29	12	11
47	NOKAD MEBU	MEBU12	J1	10/13/83	11/07/83	0.B0	17	100	50	0	0	1	0	0	1
48	NOKAD MEBU	MEBU13	J2	11/07/83	12/08/83	1.00	17	100	33	0	0	2	1	1	0
49	NOKAD MEBU	MEBU14	J3	01/28/84	02/28/84	1.00	16	100	33	0	0	1	1	0	0

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Table D-5. Systems Maintenance Block Release Raw Data

#	SUB MARKET SYSTEM	RLS ID	RLS START DATE	ENGR CUMF DATE	RLS NO.	% LEN	% PERS	PERS TOT	NO. NO.	NO. NO.	CPLX CPI X	CPLX CPI X	PRIO PRIO	PRIO PRIO	URG NORM	
					OF DED	DED	MOS.	MOS.	TYPE TYPE	TYPE TYPE	LOW ENER	HIGH MED	LOW ENER	HIGH MED	URG NORM	
50	NOKAD ME BU	MEBU	04/15/83 05/15/84	11.00	16	100	42	9	0	51	15	0	0	0	37	14
51	NOKAD ME BU	MEBU	05/14/84 08/06/84	2.70	16	100	33	0	0	4	2	0	0	0	2	0
52	NOKAD ME BU	MEBU	04/15/84 05/14/84	1.00	16	100	50	0	0	1	1	0	0	0	1	0
53	NOKAD ME BU	MEBU	11/15/82 10/15/84	11.00	16	100	47	0	0	34	18	16	0	0	25	9
54	NOKAD ME BU	MEBU	10/15/84 12/15/84	2.00	16	100	42	0	0	3	2	0	0	0	1	4
55	NOKAD ME BU	MEBU	12/15/84 01/15/85	1.00	16	100	33	0	0	1	0	1	0	0	1	0
56	NOKAD ME BU	MEBU	05/15/84 04/15/85	11.00	16	100	45	0	0	8	0	0	0	0	5	3
57	NOKAD NCS	NCS	04/05/79 03/05/80	11.00	48	100	100	0	0	245	210	35	0	0	10	112
58	NOKAD NCS	NCS	B1 03/05/80 03/07/80	0.10	48	100	5	0	0	4	4	0	0	0	0	4
59	NOKAD NCS	NCS	B2 03/07/80 04/08/80	1.00	48	100	5	0	0	1	1	0	0	0	1	0
60	NOKAD NCS	NCS	B3 04/08/80 05/16/80	1.20	48	100	50	0	0	10	5	0	0	0	5	0
61	NOKAD NCS	NCS	B4 05/16/80 05/27/80	0.30	48	100	5	0	0	2	1	1	0	0	1	0
62	NOKAD NCS	NCS	B5 05/27/80 09/16/80	3.50	48	100	5	0	0	2	1	1	0	0	1	0
63	NOKAD NCS	NCS	B6 09/16/80 02/06/81	5.00	65	100	5	0	0	2	1	0	0	0	2	0
64	NOKAD NCS	NCS	CD 10/04/80 09/04/81	11.00	65	100	44	0	0	452	371	81	0	0	2	372
65	NOKAD NCS	NCS	D1 09/04/81 11/06/81	2.00	65	100	5	0	0	1	1	0	0	0	1	0
66	NOKAD NCS	NCS	E 12/24/80 11/24/81	11.00	65	100	41	0	0	137	117	20	0	0	107	30
67	NOKAD NCS	NCS	F 05/30/81 04/30/82	11.00	67	100	47	0	0	61	44	17	0	0	36	25
68	NOKAD NCS	NCS	G 04/30/82 07/29/82	3.00	67	100	5	0	0	2	1	0	0	0	1	1
69	NOKAD NCS	NCS	H2 12/15/82 02/15/83	2.00	66	100	5	0	0	2	1	0	0	0	1	0
70	NOKAD NCS	NCS	HH1 01/15/82 12/15/82	11.00	67	100	48	0	0	77	56	21	0	0	2	35
71	NOKAD NCS	NCS	I 08/05/82 07/15/83	11.00	66	100	41	0	0	80	48	32	0	0	50	30
72	NOKAD NCS	NCS	J 11/13/82 10/13/83	11.00	66	100	41	0	0	74	41	33	0	0	49	25
73	NOKAD NCS	NCS	J1 10/13/83 11/08/83	0.80	66	100	5	0	0	6	6	0	0	0	6	0
74	NOKAD NCS	NCS	J2 11/08/83 11/28/83	0.60	66	100	5	0	0	2	1	1	0	0	1	1
75	NOKAD NCS	NCS	J3 11/28/83 01/20/84	1.80	66	100	5	0	0	1	0	1	0	0	2	5
76	NOKAD NCS	NCS	K 04/13/83 03/13/84	11.00	66	100	36	0	0	117	96	51	0	0	84	32
77	NOKAD NCS	NCS	L 03/13/84 04/30/84	1.50	66	100	5	0	0	17	13	4	0	0	11	5
78	NOKAD NCS	NCS	L2 04/30/84 06/24/84	1.90	66	100	5	0	0	3	0	3	0	0	3	0
79	NOKAD NCS	NCS	L 12/01/83 11/01/84	11.00	66	100	44	0	0	95	59	36	0	0	59	36
80	NOKAD NCS	NCS	L1 11/01/84 11/08/84	0.20	66	100	5	0	0	2	2	1	0	0	2	0
81	NOKAD NCS	NCS	L2 11/08/84 12/21/84	1.40	66	100	5	0	0	3	0	0	0	1	2	0
82	NOKAD NCS	SSC	M 05/01/84 04/01/85	11.00	66	100	47	0	0	24	11	15	0	0	15	9
83	NOKAD SSC	SSC	A 04/02/84 03/21/79	12.00	130	100	48	0	0	454	654	0	0	0	122	512
84	NOKAD SSC	SSC	A2 03/30/79 04/30/79	1.00	130	100	5	0	0	4	4	0	0	0	4	0
85	NOKAD SSC	SSC	A3 06/19/79 06/19/79	1.70	130	100	50	0	0	41	41	0	0	0	5	6
86	NOKAD SSC	SSC	A4 06/19/79 06/29/79	0.33	130	100	33	0	0	38	30	8	0	0	22	16
87	NOKAD SSC	SSC	A5 06/29/79 07/26/79	1.00	130	100	33	0	0	45	43	2	0	0	16	29
88	NOKAD SSC	SSC	A7 08/10/79 09/27/79	1.50	130	100	33	0	0	8	8	5	0	0	3	1
89	NOKAD SSC	SSC	A8 01/06/79 12/06/79	11.00	130	100	48	0	0	268	266	0	0	0	72	196
90	NOKAD SSC	SSC	B 12/06/79 01/02/80	0.90	130	100	59	0	0	18	15	0	0	0	4	14
91	NOKAD SSC	SSC	B1 01/02/80 01/14/80	0.40	130	100	5	0	0	1	0	1	0	0	0	1
92	NOKAD SSC	SSC	B2 01/14/80 02/15/80	1.00	130	100	5	0	0	3	2	1	0	0	1	1
93	NOKAD SSC	SSC	C 05/20/79 04/20/80	11.00	130	100	53	0	0	71	69	2	0	0	28	41
94	NOKAD SSC	SSC	D 01/12/81 02/12/81	1.00	130	100	53	0	0	5	5	0	0	0	14	1
95	NOKAD SSC	SSC	D1 02/01/81 04/01/81	1.00	120	100	5	0	0	2	2	0	0	0	1	2
96	NOKAD SSC	SSC	E 06/27/80 05/27/81	11.00	120	100	55	0	0	153	119	43	0	0	109	44
97	NOKAD SSC	SSC	F 09/17/80 08/17/81	11.00	120	100	36	0	0	113	103	10	0	0	72	41

Table D-5. Systems Maintenance Block Release Raw Data

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Table D-5. Systems Maintenance Block Release Raw Data

#	SOFTWARE SYSTEM	RLS ID	RLS START DATE	ENGR COMP DATE	RLS LEN OF MOS PERS	% PERS PERG	TOT NO. DED MOS S/W RLS EST	NO. TYPE	NO. CPLX CPI X	NO. CPLX PRIO	NO. NO. NO.
					MOS	PERG	ACT	CNFG CORR	HIGH MED	PRIO	URG NORM
148	WR-ALC JTIDS ASIT/ULCP	B1	10/22/84 07/01/85	8-50	10	50 100	0	0	0	0	0
149	WR-ALC JTIDS E-3A AWACS/ULCP	B1	10/22/84 07/01/85	B-50	10	50 100	0	0	0	0	0
150	WR-ALC JTIDS SF/ULCP	B1	/ / /	0.00	3	50 100	0	0	0	0	0
151	WR-ALC JTIDS SYS EXERCISER	B1	10/10/82 02/01/85	27-50	4	50 100	0	0	0	0	0
152	WR-ALC FAVE T AFSP	B1	01/01/83 09/01/85	32-00	4	70 69	0	0	0	0	0
153	WR-ALC FAVE T OFP	B2	01/01/84 06/01/86	29-00	4	70 66	0	0	0	0	0
154	WR-ALC FAVE T OFP	B1	01/01/84 06/01/86	29-00	4	70 66	0	0	0	0	0
155	SM-ALC F-111D WNC	D16	09/01/74 03/01/75	6-00	8	95 100	0	0	0	0	0
156	SM-ALC F-111D WNC	D17	09/01/75 05/01/76	8-00	8	95 100	0	0	0	0	0
157	SM-ALC F-111D WNC	D18	07/01/76 01/01/77	6-00	8	95 58	67	0	20 8	10	0
158	SM-ALC F-111D WNC	D19	08/01/76 05/01/77	9-00	8	95 78	104	0	24 12	12	0
159	SM-ALC F-111D WNC	D20	04/01/80 04/01/81	12-00	8	95 100	0	0	0	0	0
160	SM-ALC F-111F WNC	F10	05/01/75 04/01/76	11-00	7	90 100	0	0	0	0	0
161	SM-ALC F-111F WNC	F11	05/01/76 04/01/77	11-00	7	90 82	0	0	0	0	0
162	SM-ALC F-111F WNC	F12	12/01/76 12/01/77	12-00	7	90 83	119	0	46 23	23	0
163	SM-ALC F-111F WNC	F12A	06/01/78 03/01/79	9-00	7	90 100	0	0	0	0	0
164	SM ALC F-111F WNC	F13	09/01/79 03/01/80	6-00	7	90 100	0	0	0	0	0
165	SM-ALC F-111F WNC	F14	04/01/82 11/01/83	18-00	7	33 100	0	0	0	0	0
166	SM-ALC FB-111 WNC	F15	09/01/73 03/01/74	6-90	7	95 100	0	0	0	0	0
167	SM-ALC FB-111 WNC	F16	09/01/74 07/01/75	10-00	7	95 100	0	0	0	0	0
168	SM-ALC FB-111 WNC	F16A	09/01/75 06/01/76	9-00	7	95 100	0	0	0	0	0
169	SM-ALC FB-111 WNC	F16B	01/01/77 07/01/77	6-00	7	95 100	84	0	19 7	12	0
170	SM-ALC FB-111 WNC	F16C	01/01/78 01/01/79	12-00	7	95 100	103	0	25 8	17	0
171	SM-ALC FB-111 WNC	F16D	09/01/79 05/01/80	8-00	7	95 100	0	0	0	0	0
172	CASILE AFB B-52	CPI	10/07/77 11/02/80	36-00	3	45 100	0	0	0	0	0
173	CASILE AFB B-52	C1	10/02/80 09/01/81	11-00	3	45 100	0	0	0	0	0
174	CASILE AFB B-52	C1	09/01/81 09/15/82	12-50	3	45 100	0	0	0	0	0
175	CASILE AFB B-52	WSI	/ / /	0.00	40	50 100	0	0	0	0	0
176	CASILE AFB C-115	WSI	/ / /	0.00	10	50 100	0	0	0	0	0
177	CASILE AFB T-4 MR T 4	WSI	/ / /	0.00	3	45 100	0	0	0	0	0
	SIMULATOR										
178	00-ALC F-16 FCC	B155	01/01/83 12/31/84	24-00	12	80 100	0	0	0	0	0
179	00-ALC F-16 FCC	F131	01/01/83 05/31/83	3-00	12	80 90	0	0	0	0	0
180	00-ALC F-16 FCC	F132	04/30/83 09/30/83	5-00	12	80 90	0	0	0	0	0
181	00-ALC F-16 FCC	F133	10/01/83 12/31/83	3-00	12	80 100	0	0	0	0	0
182	00-ALC F-16 FCC	F134	01/01/84 03/31/84	3-00	12	80 100	0	0	0	0	0
183	00-ALC F-16 FCC	F135	04/01/84 05/31/84	2-00	12	80 100	0	0	0	0	0
184	00-ALC F-16 FCC	F136	06/01/84 07/31/84	2-00	12	80 100	0	0	0	0	0
185	00-ALC F-16 FCC	F137	08/01/84 09/01/84	1-00	12	80 100	0	0	0	0	0
186	00-ALC F-16 FCC	F04	09/01/84 09/01/84	0-05	12	80 50	0	0	0	0	0
187	00-ALC F-16 FCC	F05	09/01/84 09/01/84	0-05	12	80 50	0	0	0	0	0
188	00-ALC F-16 HU		/ / /	0.00	3	100 100	0	0	0	0	0
189	00-ALC F-16 OFT	DMD1	03/01/84 09/01/84	6-00	4	100 81	0	0	0	0	0
190	00-ALC F-16 OFT	DMD2	07/01/84 11/01/84	4-00	4	100 50	0	0	0	0	0
191	00-ALC F-16 OFT	DMD3	09/01/84 12/31/84	4-00	6	100 5	0	0	0	0	0
192	00-ALC F-16 FLK		/ / /	0.00	8	90 100	0	0	0	0	0
193	00-ALC F-16 SMS	H155	01/01/83 12/31/84	24-00	9	85 100	0	0	0	0	0
194	00-ALC F-16 SMS	SI-1	01/01/83 06/01/83	5-00	4	85 81	0	0	0	0	0
195	00-ALC F-16 SMS	SI-2	04/01/83 07/01/83	3-00	9	85 50	0	0	0	0	0

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Table D-5. Systems Maintenance Block Release Raw Data

#	SOFTWARE SYSTEM	R/S ID	R/S START DATE	ENGR CONF DATE	R/S LEN OF MOS PERS S/W	% DED MOS. ACT	TOT PERS EST	NO. TYPE	NO. TYPE	NO. TYPE	NO. CPLX CONV	NO. FRIO MED	NO. FRIO HIGH	NO. URG MED	NO. URG HIGH						
								NO. DED MOS.	NO. DED MOS.	NO. DED MOS.	NO. CHNG CORR	NO. ENH CORR	NO. LOW EMEP	NO. HIGH EMEP	NO. FRIOD	NO. FRIOH	NO. URGD	NO. URGH	NO. FRIOD	NO. FRIOH	
196	00-ALC F-16	SMS	SF 2A 06/01/83	09/30/83	4.00	9	85	88	0	0	17	13	4	0	2	8	7	0	0	1/	1
197	00-ALC F-16	SMS	SF 2B 10/27/83	11/28/83	0.05	9	85	50	0	0	3	2	1	0	2	1	1	0	0	3	3
198	00-ALC F-16	SMS	SF 3 10/01/83	12/31/83	3.00	9	85	100	0	0	22	14	8	0	3	11	8	0	0	0	22
199	00-ALC F-16	SMS	SF 3A 01/05/84	01/06/84	0.05	9	85	50	0	0	2	1	1	0	1	1	0	0	0	0	
200	00-ALC F-16	SMS	SF 4 01/01/84	02/29/84	2.00	9	85	100	0	0	9	8	1	0	2	5	2	0	0	9	
201	00-ALC F-16	SMS	SF 5 03/01/84	04/30/84	2.00	9	85	100	0	0	7	6	4	2	1	5	1	1	0	0	
202	00-ALC F-16	SMS	SF 6 05/01/84	05/31/84	1.00	9	85	100	0	0	6	4	1	1	0	1	1	0	0	6	
203	00-ALC F-16	SMS	SF 7 06/01/84	08/01/84	2.00	9	85	100	0	0	1	2	1	0	0	1	1	0	0	2	
204	00-ALC F-16	SMS	SF 05 08/01/84	08/01/84	0.05	9	85	50	0	0	1	0	0	0	0	0	0	0	0	1	
205	00-ALC F-4	MOTS	D503 08/01/83	05/01/84	9.00	2	100	100	0	0	11	9	2	0	2	4	3	0	0	11	
206	00-ALC F-4	MOTS	NEXT 11/15/84	01/01/86	13.50	0	100	100	0	0	16	13	3	0	3	9	4	0	0	16	
207	00-ALC F-4E	AN/ARN-101	1-203 01/01/83	05/01/84	16.00	6	60	100	1	0	21	17	4	0	1	12	8	0	0	21	
208	00-ALC F-4E	AN/ARN-101	NEXT 11/15/84	01/01/86	13.50	6	60	100	0	0	35	28	7	0	7	19	9	0	0	35	
209	00-ALC F-4G	AN/ARN-101	7-01 10/01/84	05/01/85	7.00	5	100	50	0	0	4	3	1	0	1	3	0	0	4	0	
210	00-ALC F-4G	AN/ARN-101	8-01 10/01/84	04/01/86	18.00	5	100	80	0	0	33	28	5	0	5	12	16	0	0	33	
211	00-ALC F-4G	LRU 1/PACM	F004 09/01/82	01/10/84	16.50	6	100	97	1	0	16	13	3	0	2	9	5	0	0	16	
212	00-ALC F-4G	LRU 1/PACM	F005 12/03/84	10/01/85	10.00	6	100	95	0	0	15	15	0	0	3	5	7	0	0	3	
213	00-ALC MINUTE WING 11/HS-2015	1	01/01/85	08/21/85	8.00	8	30	88	0	0	20	15	5	0	50	0	0	0	0	2015	
214	00-ALC MINUTE WING VI/HS-29	1	07/01/85	11/30/85	5.00	4	5	80	12	0	500	250	50	0	400	0	0	0	0	29	
215	00-ALC MINUTE WINGS/Hb-28	1	02/01/84	12/21/85	21.00	8	30	100	50	0	250	75	125	0	50	5	13	0	0	232	
216	00-ALC RF-4C	AN/ARN-101	1-203 01/01/83	05/01/84	16.00	6	40	100	0	0	26	21	5	0	0	14	10	0	0	26	
217	00-ALC RF-4C	AN/ARN-101	NEXT 11/15/84	01/01/86	13.50	6	40	100	0	0	25	20	5	0	4	15	6	0	0	25	
218	00-ALC ALCM	LIT	E430 09/01/81	02/01/82	5.00	12	100	100	1	0	60	1	0	1	0	1	0	1	0	0	
219	00-ALC ALCM	LIT	E499 11/01/82	01/01/83	2.00	18	100	100	0	0	36	37	20	7	10	0	11	26	0	30	
220	00-ALC ALCM	LIT	E513 04/01/83	12/01/83	8.00	10	100	94	200	80	28	20	5	0	7	7	0	0	20		
221	00-ALC ALCM	LIT	E525 11/01/83	01/01/85	14.00	9	100	96	294	126	38	25	11	2	1	11	26	0	0	15	
222	00-ALC ALCM	LFT	E499 01/01/82	07/01/82	6.00	11	100	100	0	0	7	4	1	2	0	3	4	0	2		
223	00-ALC ALCM	LPT	E503 05/01/82	01/01/82	8.00	9	100	100	0	0	6	5	1	2	0	5	3	0	0		
224	00-ALC ALCM	LPT	E536 08/01/84	10/01/84	2.00	7	100	100	0	0	2	1	1	0	1	0	1	0	1		
225	00-ALC ALCM	OFF	R15 08/01/81	10/01/82	14.00	10	72	100	0	0	26	24	2	0	1	0	24	0	0		
226	00-ALC ALCM	OFF	R16 03/01/82	06/01/82	3.00	10	72	100	0	0	1	1	0	0	0	0	0	0	0		
227	00-ALC ALCM	OFF	R17 01/01/83	09/01/83	8.00	10	72	100	0	0	7	5	0	2	4	3	0	0	0		
228	00-ALC ALCM	OFF	R18 09/01/84	10/01/85	13.00	10	72	100	0	0	9	4	0	0	0	0	0	0	0		
229	00-ALC B-1B	CADC	/ /	/ /	/	0.00	1	100	0	0	2	70	100	0	0	0	0	0	0	5	
230	00-ALC B-1B	BNST	H1 12/05/85	09/30/84	10.00	3	20	100	0	0	4	95	100	0	0	0	0	0	0	0	
231	00-ALC B-1B	F1SS	/ /	/ /	/	0.00	4	100	0	0	6	15	100	0	0	0	0	0	0	0	
232	00-ALC B-1B	MU-1 ExtC	/ /	/ /	/	0.00	6	50	100	0	0	0	0	0	0	0	0	0	0	0	
233	00-ALC B-1B	MU-2 ExtC	/ /	/ /	/	0.00	6	50	100	0	0	0	0	0	0	0	0	0	0	0	
234	00-ALC B-1B	INS	V17 08/01/84	03/01/85	7.00	9	100	50	0	0	1	1	0	0	0	0	0	0	0		
235	00-ALC B-1B	INS	V18 08/01/84	03/01/85	7.00	9	100	50	0	0	1	1	0	0	0	0	0	0	0		
236	00-ALC B-1B	EMUX	/ /	/ /	/	0.00	4	100	0	0	0	0	0	0	0	0	0	0	0	0	
237	00-ALC B-1B	EMUX	/ /	/ /	/	0.00	4	100	0	0	0	0	0	0	0	0	0	0	0	0	
238	00-ALC B-1B	F/GCMs	/ /	/ /	/	0.00	4	100	0	0	0	0	0	0	0	0	0	0	0	0	
239	00-ALC B-1B	INS	/ /	/ /	/	0.00	1	100	0	0	0	0	0	0	0	0	0	0	0	0	
240	00-ALC B-1B	INS	/ /	/ /	/	0.00	2	100	0	0	0	0	0	0	0	0	0	0	0	0	
241	00-ALC E-1A	OMEGA	H1 03/01/83	10/01/83	7.00	0	100	50	0	0	0	0	0	0	0	0	0	0	0	0	
242	00-ALC E-1A	OMEGA	B2 12/01/84	12/01/85	12.00	0	100	100	0	0	0	0	0	0	0	0	0	0	0	0	
243	00-ALC E-1A	SMCF	H1 03/07/84	07/17/84	12.00	4	20	100	0	0	0	0	0	0	0	0	0	0	0	0	
244	00-ALC E-1A	SMCF	H2 10/29/84	02/21/85	4.00	1	100	100	0	0	0	0	0	0	0	0	0	0	0	0	

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Table D 5. Systems Maintenance Block Release Raw Data

#	SOFWARE SYSTEM	RLS ID	RLS START DATE	ENGR COMP DATE	RLS LEN OF PERS	% PERS	TOT MOS.	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE	NO. TYPE		
					MOS. PERS	S/W RLS EST	ACT	CHNG CORR	ENH CONV	HIGH MED	LOW EMER	URG	NORM	CPLX	CPLX	CPLX	CPLX	CPLX		
245	OC-ALC E-3A	SRCP	CB1	01/31/84	01/31/85	12.00	11	40	50	0	0	0	0	0	0	0	0	0		
246	OC-ALC E-3A	SRCP	MB1	01/31/84	01/31/85	12.00	11	60	50	0	0	0	0	0	0	0	0	0		
247	OC-ALC E-3A	SAGSCF	B1	06/01/84	07/01/85	13.00	3	50	100	7	11	4	2	0	0	0	0	0		
248	OC-ALC GLCM	DFS		/	/	0.00	2	80	100	0	0	0	0	0	0	0	0	0		
249	OC-ALC GLCM	M-DID		/	/	0.00	7	100	100	0	0	0	0	0	0	0	0	0		
250	OC-ALC GLCM	NP-T		/	/	0.00	6	50	100	0	0	0	0	0	0	0	0	0		
251	OC-ALC GLCM	OFP		/	/	0.00	3	100	100	0	0	0	0	0	0	0	0	0		
252	OC-ALC GLCM	WCS		/	/	0.00	4	50	100	0	0	0	0	0	0	0	0	0		
253	OC-ALC SRAM		B1	06/01/78	01/01/79	7.00	15	75	5	0	1	0	0	0	0	0	0	0		
254	OC-ALC SRAM	OFF	B2	02/01/78	06/01/79	16.00	15	100	0	0	19	0	19	0	0	0	0	0	0	
255	TINKER AFB E-3A	AUCP	10A	10/01/77	02/15/78	4.50	67	50	100	0	27	24	3	0	1	26	0	0	0	
256	TINKER AFB E-3A	AUCP	11B	02/15/78	08/01/78	5.50	67	50	100	0	0	88	14	0	2	84	0	0	0	
257	TINKER AFB E-3A	AUCP	10C	08/01/78	01/02/79	5.00	67	50	100	0	0	77	63	14	0	1	4	0	0	
258	TINKER AFB E-3A	AULF	20A	06/01/79	12/01/79	6.00	67	50	100	0	76	66	10	0	2	6	68	0	0	
259	TINKER AFB E-3A	AULF	20AA	12/01/79	02/04/80	2.00	67	50	100	0	50	49	1	0	8	42	0	0	0	
260	TINKER AFB E-3A	AULF	20B	02/04/80	06/01/80	4.00	67	50	100	0	0	41	37	4	0	1	28	0	0	
261	TINKER AFB E-3A	AULF	21A	06/01/80	02/01/81	8.00	67	50	100	0	0	51	34	17	0	3	10	38	0	0
262	TINKER AFB E-3A	AUCP	21A1	02/01/81	05/19/81	1.50	67	50	100	0	0	5	5	0	0	0	4	1	0	
263	TINKER AFB E-3A	AUCP	22A	05/19/81	12/22/81	7.00	67	50	100	0	0	99	78	21	0	4	15	80	0	0
264	TINKER AFB E-3A	AUCP	22A1	12/22/81	04/30/82	4.00	67	50	100	0	0	7	7	0	0	2	5	0	0	
265	TINKER AFB E-3A	AUCP	22B	04/30/82	08/23/82	3.50	67	50	100	0	0	71	68	3	0	2	17	52	0	0
266	TINKER AFB E-3A	AUCP	22B1	08/23/82	01/03/83	4.50	67	50	100	0	0	14	14	0	0	0	7	0	0	
267	TINKER AFB E-3A	AUCP	22B2	07/01/83	12/09/83	5.00	67	50	100	0	0	34	33	1	0	1	33	0	0	
268	TINKER AFB E-3A	AUCP	23A	12/09/83	07/25/84	7.50	67	50	100	0	0	134	117	17	0	4	21	109	0	0
269	TINKER AFB E-3A	AUCP	24A	01/22/85	09/22/85	8.00	67	50	69	0	0	57	42	15	0	2	11	44	0	0
270	TINKER AFB E-3A	AUCP	24A1	04/22/85	12/22/85	8.00	67	50	69	0	0	47	35	12	0	1	1	45	0	0
271	TINKER AFB E-3A	AUCP	20A	07/01/83	08/15/83	1.50	67	50	95	0	0	14	12	2	0	0	0	14	0	0
272	TINKER AFB E-3A	AUCP	30A1	08/15/83	12/09/83	4.00	67	50	95	0	0	34	32	2	0	0	2	32	0	0
273	TINKER AFB E-3A	AUCP	51A	07/25/84	10/03/84	2.00	67	50	5	0	0	1	0	1	0	0	0	1	0	
274	TINKER AFB E-3A	JLILYFS	10A	10/01/77	02/15/78	4.50	46	50	100	0	0	10	3	7	0	0	1	9	0	0
275	TINKER AFB E-3A	JLILYFS	10B	02/15/78	08/01/78	5.50	46	50	100	0	0	14	7	0	0	0	1	13	0	0
276	TINKER AFB E-3A	UTILITIES	10C	08/01/78	01/02/79	5.00	46	50	100	0	0	21	11	10	0	1	1	19	0	0
277	TINKER AFB E-3A	UTILITIES	20A	06/01/79	12/09/82	6.00	46	50	100	0	0	51	33	18	0	0	0	44	0	0
278	TINKER AFB E-3A	UTILITIES	20B	02/04/80	06/01/80	4.00	46	50	100	0	0	25	19	6	0	1	2	22	0	0
279	TINKER AFB E-3A	UTILITIES	21A	06/01/80	02/01/81	8.00	46	50	100	0	0	43	22	21	0	0	2	9	0	0
280	TINKER AFB E-3A	UTILITIES	22A	05/19/81	12/22/81	7.00	46	50	100	0	0	39	27	12	0	2	6	31	0	0
281	TINKER AFB E-3A	UTILITIES	22A1	12/22/81	04/30/82	4.00	46	50	5	0	0	1	0	0	0	0	0	0	0	
282	TINKER AFB E-3A	UTILITIES	22B	04/30/82	08/23/82	3.50	46	50	100	0	0	24	21	3	0	1	7	16	0	0
283	TINKER AFB E-3A	UTILITIES	22B2	07/01/83	12/09/83	5.00	46	50	5	0	0	1	1	0	0	0	0	0	0	
284	TINKER AFB E-3A	UTILITIES	23A	12/09/83	07/25/84	7.50	46	50	100	0	0	25	18	7	0	1	6	18	0	0
285	TINKER AFB E-3A	UTILITIES	24A	01/22/85	09/22/85	8.00	46	50	50	0	0	27	17	10	0	1	5	21	0	0
286	TINKER AFB E-3A	UTILITIES	25A	04/22/85	12/22/85	8.00	46	50	5	0	0	1	1	0	0	0	1	1	0	
287	TINKER AFB E-3A	UTILITIES	30A	07/01/83	08/15/83	1.50	46	50	95	0	0	25	16	9	0	1	1	23	0	0
288	LANGLEY JTIDS	AS11/F1OCF	V1	06/22/84	04/25/85	10.00	5	80	100	0	0	5	1	4	0	0	3	2	0	0
289	LANGLEY STRTS	STRTS	V1	/	/	0.00	0	1	0	0	0	0	0	0	0	0	0	0	0	0
290	LANGLEY TACS	CAMS	1.0	01/01/81	05/11/81	5.00	50	100	0	0	0	0	0	0	0	0	132	0	0	132
291	LANGLEY TACS	CAMS	1.1	09/01/81	09/30/81	1.00	50	100	0	0	0	0	0	0	0	0	2	0	0	2
292	LANGLEY TACS	CAMS	1.2	09/30/81	12/14/81	2.50	50	100	0	0	0	0	0	0	0	0	21	0	0	85
293	LANGLEY TACS	CAMS	1.3	12/14/81	03/01/82	2.50	50	100	0	0	0	0	0	0	0	0	19	3	0	11

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Table D-5. Systems Maintenance Block Release Raw Data

#	SOFTWART SYSTEM	RLS ID	RLS START DATE	ENGR COMP	RLS DATE	LEN	% PERS	TOT NO.	NO. OF DED MOS.	NO. DED MOD.	NO. TYPE	CPI X	CPLX	PRIO	PRIO	PRIO	PRIO	PRIO	NO. CFI X	NO. CFI X	NO. CFI X	NO. URG	NO. URG	NO. URG	
								S/W	RLS	EST	ACT	CHNG	CONV	ENH	LOW	MED	HIGH	HIGH	TYPE	TYPE	TYPE	EMER			
294	LANGLEY TACS	2-4	03/01/82	09/15/82	6-50	50	100	100	0	0	0	66	42	24	0	29	12	25	0	0	0	0	66	0	
295	LANGLEY TACS	3-0	09/15/82	10/20/83	13.00	43	100	100	0	0	0	35	12	23	0	9	16	10	0	0	0	0	35	0	
296	LANGLEY TACS	3-1	10/20/83	08/03/84	9.50	36	100	100	0	0	0	35	25	19	0	6	16	13	0	0	0	0	35	0	
297	LANGLEY TACS	4-0	08/03/84	07/15/85	11.50	29	100	100	0	0	0	27	11	16	0	10	14	3	0	0	0	0	27	0	
298	LANGLEY TIP1	1	10/01/74	09/01/75	11.00	55	100	100	0	0	0	55	55	0	0	6	49	0	0	0	0	0	0	55	0
299	LANGLEY TIP1	1C	09/01/75	05/01/76	8.00	55	100	100	0	0	0	103	91	12	0	17	62	24	0	0	0	0	103	0	
300	LANGLEY TIP1	2B	05/01/76	04/01/77	11.00	54	100	100	0	0	0	235	215	20	0	33	136	66	0	0	0	0	235	0	
301	LANGLEY TIP1	2C	04/01/77	10/01/77	6.00	54	100	100	0	0	0	222	207	15	0	27	124	71	0	0	0	0	222	0	
302	LANGLEY TIP1	2D	10/01/77	05/01/78	7.00	55	100	100	0	0	0	173	160	13	0	26	87	61	0	0	0	0	173	0	
303	LANGLEY TIP1	2E	05/01/78	10/01/78	5.00	56	100	100	0	0	0	251	251	0	0	15	143	93	0	0	0	0	251	0	
304	LANGLEY TIP1	3A	10/01/78	04/01/79	6.00	48	100	100	0	0	0	230	224	6	0	21	106	103	0	0	0	0	230	0	
305	LANGLEY TIP1	3B	04/01/79	08/01/79	4.00	49	100	100	0	0	0	120	108	12	0	17	71	32	0	0	0	0	120	0	
306	LANGLEY TIP1	3C	08/01/79	03/01/80	7.00	45	100	100	0	0	0	166	151	15	0	18	96	52	0	0	0	0	166	0	
307	LANGLEY TIP1	3D	03/01/80	08/01/80	5.00	45	100	100	0	0	0	97	94	5	0	8	68	21	0	0	0	0	97	0	
308	LANGLEY TIP1	4A	08/01/80	03/01/81	7.00	45	100	100	0	0	0	105	97	6	0	15	55	36	0	0	0	0	105	0	
309	LANGLEY TIP1	4B	03/01/81	09/01/81	6.00	46	100	100	0	0	0	150	140	10	0	18	81	51	0	0	0	0	150	0	
310	LANGLEY TIP1	4C	09/01/81	01/01/82	4.00	43	100	100	0	0	0	87	78	9	0	13	45	29	0	0	0	0	87	0	
311	LANGLEY TIP1	83	01/01/82	01/01/83	12.00	43	100	100	0	0	0	195	176	19	0	29	115	61	0	0	0	0	195	0	
312	LANGLEY TIP1	84	01/01/82	01/01/84	12.00	43	100	100	0	0	0	214	188	26	0	54	128	32	0	0	0	0	214	0	
313	LANGLEY TIP1	85	01/01/84	01/01/85	12.00	45	100	100	0	0	0	121	98	23	0	34	76	11	0	0	0	0	121	0	
314	LANGLEY TIP1	11A	03/01/83	01/01/84	10.00	16	85	100	0	0	0	12	83	63	0	20	4	25	54	1	0	0	0	82	0
315	LANGLEY TIP1	12	03/01/84	01/01/85	10.00	16	85	100	0	0	0	20	105	76	29	0	4	32	69	1	0	0	0	104	0
316	LANGLEY TIP1	12-1	03/01/85	06/01/85	3.00	16	85	100	0	0	7	13	11	2	0	2	4	7	0	0	0	0	13	0	
317	LM/MARRES/TEREC	9A	10/01/81	01/01/83	15.00	16	85	100	0	0	18	268	148	60	0	13	31	134	1	0	0	0	207	0	
318	LANGLEY 407L	HUGHES UTIL	V4-3	02/01/80	06/09/80	4.50	2	10	100	0	0	0	14	6	8	0	0	0	0	0	0	0	0	14	0
319	LANGLEY 407L	HUGHES UTIL	V4-4	07/01/80	06/08/81	11.00	2	10	100	0	0	0	8	6	2	0	9	0	0	0	0	0	0	9	0
320	LANGLEY 407L	HUGHES UTIL	V4-5	06/01/81	08/02/82	14.00	2	10	100	0	0	0	29	20	9	0	2	29	0	0	0	0	29	0	
321	LANGLEY 407L	HUGHES UTIL	V4-6	07/01/82	11/30/82	5.00	2	10	100	0	0	0	13	10	3	0	0	13	0	0	0	0	13	0	
322	LANGLEY 407L	HUGHES UTIL	V4-8	11/01/83	06/01/84	7.00	2	10	100	0	0	0	40	21	19	0	0	40	0	0	0	0	40	0	
323	LANGLEY 407L	HUGHES UTIL	V4-9	07/01/84	12/01/84	5.00	2	10	100	0	0	0	1	1	0	0	0	1	0	0	0	0	1	0	
324	LANGLEY 407L	HUGHES UTIL	V5-0	12/01/84	07/15/85	7.00	2	10	100	0	0	0	16	15	1	0	0	16	0	0	0	0	16	0	
325	LANGLEY 407L	1MF UTIL	V4-1	/	/	0.00	5	80	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0		
326	LANGLEY 407L	1MF UTIL	V4-2	02/01/79	08/01/79	6.00	40	60	100	0	0	0	40	26	14	0	10	16	14	0	0	0	0	30	0
327	LANGLEY 407L	1MF UTIL	V4-3	09/01/79	01/21/80	4.50	40	60	100	0	0	0	43	27	16	0	5	24	14	0	0	0	0	38	0
328	LANGLEY 407L	1MF UTIL	V4-4	02/01/80	06/09/80	4.50	40	60	100	0	0	0	88	61	27	0	25	27	36	0	0	0	0	74	0
329	LANGLEY 407L	1MF UTIL	V4-5	07/01/80	06/08/81	11.00	40	60	100	0	0	0	153	125	28	0	10	98	105	0	0	0	0	137	0
330	LANGLEY 407L	1MF UTIL	V4-6	06/01/81	08/02/82	14.00	40	60	100	0	0	0	90	61	29	0	15	23	42	0	0	0	0	71	0
331	LANGLEY 407L	1MF UTIL	V4-7	07/01/82	11/39/82	5.00	40	60	100	0	0	0	45	30	15	0	7	16	22	0	0	0	0	39	0
332	LANGLEY 407L	1MF UTIL	V4-8	11/01/82	01/31/83	11.00	40	60	100	0	0	0	81	54	27	0	11	18	32	0	0	0	0	68	0
333	LANGLEY 407L	1MF UTIL	V4-9	07/01/84	06/01/84	7.00	40	60	100	0	0	0	39	24	15	0	13	19	19	0	0	0	0	27	0
334	LANGLEY 407L	1MF UTIL	V5-0	12/15/84	07/15/85	7.00	40	60	100	0	0	0	65	55	10	0	11	15	15	0	0	0	0	51	0
335	LANGLEY 407L	1MF UTIL	V5-1	07/15/85	12/15/85	5.00	40	60	100	0	0	0	79	68	11	0	13	35	0	0	0	0	52	0	

## **E. Systems Descriptions**

APPENDIX E

SYSTEMS DESCRIPTIONS

Because it is highly probable that all the software systems reviewed in this study will not be familiar to the reader, this appendix contains systems descriptions for most systems listed in table E-1. For some systems, appropriate descriptions were not available to the authors.

Table E-1  
Software Systems Examined

<u>SITE</u>	<u>SYSTEM</u>	<u>SOFTWARE SYSTEM</u>
NORAD	CSS	CSS
NORAD	MDS	MDS
NORAD	MEBU	MEBU
NORAD	NCS	NCS
NORAD	SSC	SSC
WR-ALC	ALR-46	ALR-46
WR-ALC	ALR-69	ALR-69
WR-ALC	AN/ALQ-131	AGEOP
WR-ALC	AN/ALQ-131	BTG
WR-ALC	AN/ALQ-131	OFP
WR-ALC	ALQ-131	UUT
WR-ALC	APR-38	APR-38
WR-ALC	B-52 EVS ATE	ASQ-151
WR-ALC	E-3A AVIONICS ATE	AN/GSM-285(B)
WR-ALC	E-3A AVIONICS ATE	AN/GSM-285(W)
WR-ALC	F-15	CC
WR-ALC	F-15	RADAR
WR-ALC	F-15 AVIONICS ATE	ADTS,AIS
WR-ALC	JTIDS	ASIT/OCP
WR-ALC	JTIDS	E-3A AWACS/OCP
WR-ALC	JTIDS	SP/USER
WR-ALC	JTIDS	SYS EXERCISER
WR-ALC	PAVE TACK	AISF
WR-ALC	PAVE TACK	OFP
SM-ALC	F-111D	WEAP-NAV COMPUTER
SM-ALC	F-111F	WEAP-NAV COMPUTER
SM-ALC	FB-111A	WEAP-NAV COMPUTER
CASTLE AFB	A T-4	A T-4 SIMULATOR
CASTLE AFB	B-52	CPT
CASTLE AFB	B-52	WST
CASTLE AFB	KC-135	WST
OO-ALC	F-16	FCC
OO-ALC	F-16	HUD
OO-ALC	F-16	OFT
OO-ALC	F-16	FCR
OO-ALC	F-16	SMU
OO-ALC	F-4	MDTS
OO-ALC	F-4E	AN/ARN-101
OO-ALC	F-4G	AN/ARN-101
OO-ALC	F-4G	LRU-1/ACM
OO-ALC	MINUTEMAN	WING II/2015
OO-ALC	MINUTEMAN	WING VI/HS-29
OO-ALC	MINUTEMAN	WINGS/HS-28

Table E-1  
Software Systems Examined (Continued)

CO-ALC	MINUTEMAN II	SSAS/CAPS
OO-ALC	MINUTEMAN II	WING V/HEG/RATS
OO-ALC	MINUTEMAN II	WING VI/HEG/RATS
OO-ALC	RF-4	CAN/ARN-101
OC-ALC	ALCM	LEVEL 1 TEST
OC-ALC	ALCM	LOADED PYLON TEST
OC-ALC	ALCM	OFP
OC-ALC	B-1B	CADC
OC-ALC	B-1B	CITS
OC-ALC	B-1B.	EMUX
OC-ALC	B-1B	F/CGMS
OC-ALC	B-1B	INS
OC-ALC	B-1B	ORS
OC-ALC	B-52	BNST
OC-ALC	B-52	FTSS
OC-ALC	B-52	MC-1 EXEC
OO-ALC	B-52	MC-2 EXEC
OC-ALC	E-3	AINS
OC-ALC	E-3A	OMEGA
OC-ALC	E-3A	SMCP
OC-ALC	E-3A	SRCP
OC-ALC	E-3A	SRGSCP
OC-ALC	GLCM	DPS
OC-ALC	GLCM	M-DTD
OC-ALC	GLCM	MPT
OC-ALC	GLCM	OFP
OC-ALC	GLCM	WCS
OC-ALC	SRAM	OFP
TINKER	E-3A	AOCP
TINKER	E-3A	UTIL SUPP S/W
ANGLEY	JTIDS	ASIT/TPOCP
ANGLEY	STRTS	STRTS
ANGLEY	TACS	CAFMS
ANGLEY	TIPI	DC/SR
ANGLEY	TIPI	II/MARRES/TEREC
ANGLEY	407L	HUGHES UTIL
ANGLEY	407L	TBM UTIL
ANGLEY	407L	IOPP/IMPP

ID : 1  
SITE : NORAD  
SYSTEM : CSS  
SOFTWARE SUBSYSTEM : CSS  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The NORAD Cheyenne Mountain Complex (NCMC) software is a complex of communications-electronics systems for space surveillance, missile warning, and related communications and support functions. The five major software subsystems include NCS, SSC, CSS, MEEU, and MDS.

The Communications System Segment (CSS) provides the required communications interfaces between program elements, between NCMC systems and external systems. It consists of Honeywell Information System (HIS) and NUVA digital computers and application software/firmware.

ID : 2  
SITE : NORAD  
SYSTEM : MDS  
SOFTWARE SUBSYSTEM : MDS  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The NORAD Cheyenne Mountain Complex (NCCM) software is a complex of communications-electronics systems for space surveillance, missile warning, and related communications and support functions. The five major software subsystems include NCS, SSC, CSG, MEBU, and MDS.

The Modular Display Sub-system (MDS) consists of Raytheon consoles, Data General NOVA Digital Computers, Ford Aerospace and Communications Corporation (FACC) developed hardware and joint FACC and System Development Corporation developed software/firmware. It provides the NORAD Computer System (NCS) and the Space Surveillance Center (SSC) with the required man-machine interface with the Embedded Computer Resources in the NCCM. Additionally, a CINCNORAD Remote Display Information Terminal has been established which provides CINCNORAD on-line access to NCS data.

ID : 7  
SITE : NORAD  
SYSTEM : MEBU  
SOFTWARE SUBSYSTEM : MEBU  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The NORAD Cheyenne Mountain Complex (NCCMC) software is a complex of communications-electronics systems for space surveillance, missile warning, and related communications and support functions. The five major software subsystems include NCS, CSC, CSS, MEBU, and MDS.

The Mission Essential Backup/Command Center Processing and Display System (MEBU/CCFOS), consists of UNIVAC 1100 series digital computers and associated display hardware, UNIVAC standard software, MEBU unique software, and CCPLS software. The MEBU provides backup to the NORAD Computer System (NCS) for the Missile Warning Mission of NORAD.

ID : 4  
SITE : NORAD  
SYSTEM : NCS  
SOFTWARE SUBSYSTEM : NCS  
SOFTWARE SUBSYSTEM TYPE : C-8  
DESCRIPTION:

The NORAD Cheyenne Mountain Complex (NCCMC) software is a complex of communications-electronics systems for space surveillance, missile warning, and related communications and support functions. The five major software subsystems include NCS, SBC, CSC, MBSU, and RBS.

The NORAD Computer System (NCS) consists of the world wide Military Command and Control System (WWMCCS), Honeywell Information Systems (HIS) 8080 digital computers and associated display hardware, related WWMCCS standard software and applications software. The NCS provides CINCNORAD with the required computer resources for command and control of the NORAD forces and for missile warning operations.

The NCS Operations programs comprise a major portion of the NCS software. These programs perform the calculations and data manipulations directly associated with aerospace defense and warning, generate and process simulated data in real-time, and record operational data. They accept real-time inputs from operators and from data sources external to NORAD and generate real-time outputs for internal display and transmission to other users.

The NCS Support software performs the calculations and data manipulations directly associated with exercise generation and data reduction. Its inputs are primarily cards, tape files and disk files, and it generates tape and disk files for use in the Operations and Utility UFCI and printed reports for use by Operations and Programming personnel.

The NCS Utility software comprises a minor portion of the NCS. This software performs the utility functions of production testing and support of the NCS Operations and NCS Support programs. It accepts real-time and non-real-time inputs from operators and generates outputs for internal display, construction, parameter assembly testing, and library routines.

## THE BDM CORPORATION

(C : 3  
 SITE : NORAD  
 SYSTEM : SSC  
 SOFTWARE SUBSYSTEM : SSC  
 SOFTWARE SUBSYSTEM TYPE : C-E  
 DESCRIPTION:

The NORAD Cheyenne Mountain Complex (NCCMC) software is a complex of communications-electronics systems for space surveillance, missile warning, and related communications and support functions. The five major software subsystems include NCS, GSC, CSS, MEGU, and MSG.

The following SSC functional areas are defined in terms of the capabilities required to support that area of the mission.

- (1) Astrodynamical Support provides for accurate, precise, rapid astrodynamical computations.
- (2) Operations Center Control supports command direction and control of SSC processing.
- (3) Automatic Catalog Maintenance satisfies requirements for automatically initiated (data triggered) processing of sensor observations, and correction and transmission of satellite orbital elements.
- (4) Launch Processing provides for detection of new satellite launches, generation and maintenance of orbital elements of new satellites, and control and efficient use of the sensor network during the new-launch time frame.
- (5) Breakup Processing provides for efficient generation, correction, maintenance, and cataloging of orbital elements of satellites associated with breakups.
- (6) Maneuver Processing provides for detection and analysis of satellite maneuvers, and for generation of elements for newly-changed satellite orbits.
- (7) Sensor Control provides for monitoring sensor environment and performance status, managing routine sensor data collection, and optimizing use of the sensor network.
- (8) Manual Analysis provides for manually-initiated selection of observations and evaluation of orbital parameters.
- (9) External Data Products provides for generation of satellite element catalog and products for use by outside system sensors and catalog data users.
- (10) Special Mission Products provides data for use in need individualized support.
- (11) Orbital Analyst Products provides for analysis of orbital space events, and for generation of data used to generate and describing satellite orbital element data.
- (12) Management Products supports management of all statistical resources on the system, and provides for system-wide statistical reporting.
- (13) System Support provides for implementation of system-wide fault detection and recovery mechanisms.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 9  
SITE : WR-ALC  
SYSTEM : ALR-40  
SOFTWARE SUBSYSTEM : ALR-40  
SOFTWARE SUBSYSTEM TYPE : EW  
DESCRIPTION:

ALR-40 is a threat warning system. The threat warning system is a software programmable radar warning set which alerts the pilot to the presence of signals emanating from threat radars. Identity and relative bearing of each threat are presented to the pilot to enable the options of avoidance and/or use of countermeasures.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 7  
SITE : WR-ALC  
SYSTEM : ALR-69  
SOFTWARE SUBSYSTEM : ALR-69  
SOFTWARE SUBSYSTEM TYPE : EW  
DESCRIPTION:

ALR-69 is a threat warning system. The threat warning system is a software programmable radar warning set which alerts the pilot to the presence of signals emanating from threat radars. Identity and relative bearing of each threat are presented to the pilot to enable the options of avoidance and/or use of countermeasures.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 3  
SITE : WR-ALC  
SYSTEM : AN/ALQ-171  
SOFTWARE SUBSYSTEM : AGEOP  
SOFTWARE SUBSYSTEM TYPE : EW  
DESCRIPTION:

The Aerospace Ground Equipment Operating System (AGEOP) is resident on the AN/ALM-186 and provides initialization of the AN/ALM-186, task scheduler, disk file manager, software drivers for I/O devices, and command interpretation. It also provides the operator interface to CUT software through a test executive program.

THE BDM CORPORATION

BDM/A-85-0510-TR

10 :  
TITLE : WR-ALE  
SYSTEM : ANA-HUD-131  
SOFTWARE SUBSYSTEM : BTG  
SOFTWARE SUBSYSTEM TYPE : EW  
DESCRIPTION:

The Blue Tape Generator (BTG) is an interactive program to generate the data required by the ECM system Operational Flight Program (OFP). This data consists of both mission data files, threat definitions and jamming techniques, and aircraft data files, i.e. amounts of prime power available. The input to the BTG is alpha-numeric text and the output is binary data tables.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 10  
SITE : WF-HALC  
SYSTEM : AN/ALQ-131  
SOFTWARE SUBSYSTEM : OFP  
SOFTWARE SUBSYSTEM TYPE : OFP  
DESCRIPTION:

The Operational Flight Program (OFP) provides control, fault detection, and hardware resource allocation during flight operations. These functions are based on data which is the output of the Blue Tape Generator and control commands from the cockpit.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 4  
SITE : NR-HL0  
SYSTEM : AN/HLD-131  
SOFTWARE SUBSYSTEM : UUT  
SOFTWARE SUBSYSTEM TYPE : ATE  
DESCRIPTION:

The Unit Under Test (UUT) software is resident on the AN/HLD-130 and provides the interface among the support equipment computer, UUT (AN/HLD-131) and the maintenance technician in the field shop. This software provides some limited automatic test capability and manual test instructions to the technician for system checkout and fault isolation.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 12  
SITE : AFM-ALC  
SYSTEM : AFR-18  
SOFTWARE SUBSYSTEM : AFR-18  
SOFTWARE SUBSYSTEM TYPE : EW  
DESCRIPTION:

The AFR-18 Homing and Warning system is comprised of four subsystems having a total of 20 Line Replaceable Units. The primary subsystems include: Receiver Set, Control Indicator Set, Homing and Warning Computer, and Computing and Optical Sight System.

THE BDM CORPORATION

BDM/A-85-0510-TR

E : 17  
BITS : 32-BIT  
SYSTEM : B-32 EVS ATE  
SOFTWARE SUBSYSTEM : AGC-151  
SOFTWARE SUBSYSTEM TYPE : ATE  
DESCRIPTION:

Not Available.

REF : 14  
SITE : WR-ALE  
SYSTEM : E-3A AVIONICS ATE  
SOFTWARE SUBSYSTEM : AN/GEM-185 (B)  
SOFTWARE SUBSYSTEM TYPE : A/E  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (CC) center in a modified Boeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs including programs for Automatic Test Equipment (ATE).

ATE provides two major capabilities: to determine rapidly whether or not the Unit Under Test (UUT) needs repair and to isolate the fault in a failed UUT to a lower level. ATE software is used for system control, translation, checkout, and execution of test programs to provide ATE Diagnosis/Fault Isolation and Support Software maintenance.

The AN/GEM-185 is a general purpose, computer-controlled Automatic Test System composed of Boeing (B) developed software and Warner Robins (W) developed software. It is used at the depot level in conjunction with adapters, test programs, and procedures for functional testing and fault isolation of digital and analog/hybrid E-3 Avionics circuit cards. In addition, it can be used for on-line generation, editing, and validation of test programs. Basically, the AN/GEM-185 provides power and signal stimuli, measurement, and signal switching to and from the UUT. Through the test program, the computer is used to control and sequence test, to generate UUT stimulus waveforms, and to analyze UUT response signals.

IE : 15

SITE : WRT-4LC

SYSTEM : E-3A AVIONICS ATE

SOFTWARE SUBSYSTEM : AN/GSM-285 (W)

SOFTWARE SUBSYSTEM TYPE : ATE

DESCRIPTION:

The E-3 Airborne warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (CCC) center in a modified Boeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

ATE provides two major capabilities: to determine rapidly whether or not the Unit Under Test (UUT) needs repair and to isolate the fault in a failed UUT to a lower level. ATE software is used for system control, translation, checkout, and execution of test programs to provide ATE Diagnosis/Fault Isolation and Support Software Maintenance.

The AN/GAM-285 is a general purpose, computer-controlled Automatic Test System composed of Boeing (B) developed software and Warner Robins (W) developed software. It is used at the circuit level in conjunction with adapters, test programs, and procedures for functional testing and fault isolation of digital and analog/hybrid E-3 Avionics circuit cards. In addition, it can be used for on-line generation, editing, and verification of test programs. Basically, the AN/GAM-285 provides power and signal stimuli, measurement, and signal switching to and from the UUT. Through the test program, the computer is used to control and sequence test, to generate UUT stimulus waveforms, and to analyze UUT response signals.

ID : 15  
SITE : WR-HLC  
SYSTEM : F-15  
SOFTWARE SUBSYSTEM : CC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The F-15, designed and manufactured by McDonnell Aircraft Company (McAire), is a single seat, twin turbofan, air superiority fighter weighing about 40,000 pounds with engines which develop approximately 25,000 pounds of thrust each. It is in the Mach 2.5 class. Armament includes 4 AIM-7 Sparrows, 4 AIM-9 Sidewinders, and a 20mm M-61 gun. The primary mission of the F-15 is air-to-air combat with ground attack as a secondary capability.

The F-15 Central Computer (CC) is an IBM developed general purpose, stored program, simplex, high speed, digital machine designated the AP-1. The CU memory is random access, non-volatile core with a capacity of 16,384 34-bit words (2 parity) which is expandable to 24,576 words.

The F-15 CC Operational Flight Program (OFF) is divided into eight program modules which primarily perform mission oriented calculations and output the results to the appropriate F-15 subsystems. The eight program modules are: Executive, Air-to-Air, Air-to-Ground, Navigation, Flight Director, Control and Display, Computer Self Test, and Math Subroutine.

ID : 17  
SITE : WR-ALC  
SYSTEM : F-15  
SOFTWARE SUBSYSTEM : RADAR  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The F-15, designed and manufactured by McDonnell Aircraft Company (McAir), is a single seat, twin turbofan, air superiority fighter weighing about 40,000 pounds with engines which develop approximately 25,000 pounds of thrust each. It is in the Mach 2.5 class. Armament includes 4 AIM-7 Sparrows, 4 ALM-9 Sidewinders, and a 20mm M-61 gun. The primary mission of the F-15 is air-to-air combat with ground attack as a secondary capability.

The radar system consists of Radar Set AN/APG-63 and Indicator Group DD-60/A. It is a coherent, X-band, multiple PRF, multi-mode, attack radar. The system searches for, acquires, and tracks airborne targets while providing a clutter-free display of all radar information. The system also provides air-to-ground mapping and ranging, as well as a radar beacon mode in both air-to-air and air-to-ground operation.

The Radar Data Processor (RDP) is a Hughes developed general purpose computer which provides the focal point for radar set operation as well as for interface with other avionics equipment. The RDP consists of a processor, a special input/output unit and integrated power supply. Three RDP configurations are planned: a 16K device using core memory, a 24K device using solid state memory, and a larger 40K solid state device to include the Programmable Signal Processor (PSP) Line Replaceable Unit as well as expansion space for the RDP.

The RDP and PSP software are loaded together. The RDP programs provide for radar acquisition, track and built-in-test functions. The PSP programs provide digital processing of the radar returns.

ID : 18  
SITE : WR-ALC  
SYSTEM : F-15 AVIONICS ATE  
SOFTWARE SUBSYSTEM : ADTS,AIS  
SOFTWARE SUBSYSTEM TYPE : ATE  
DESCRIPTION:

Automatic Test Equipment (ATE) software applies collectively to three categories: Test Software, Support Software and Control Software. Test Software includes programs which control the testing operations and procedures (including certification and fault isolation) of the ATE, and programs used to control the stimulus and measurement parameters used in testing the Avionics and ATE Unit Under Test (UUT). Support Software includes programs which aid in preparing, analyzing, and maintaining test software. This software includes ATE compilers, translation/analysis programs, and punch/print programs. Control Software includes programs used during execution of a test program which controls the nontesting operations of the ATE. This software is used to execute a test procedure but does not contain any of the stimuli or measurement parameters used in testing the UUT.

ATE system software primarily includes the Avionics Intermediate Shop (AIS) and Avionic Depot Test Station (ADTS) subsystems. The F-15 AIS ATE support software uses F-15 Adapted ATLAS as the source language, and extensively modified version of the PLACE Compiler, which is called the F-15 Adapted PLACE ATLAS (FAFA) compiler. The control language is Bendix Assembly and the control computer is the Bendix 6100. The executive/OF system controls UUT test operation, self test, CA/FI tests and mass storage edit functions. The F-15 ADTS Computer Control System (OF system) is divided into five basic functions as follows: (1) Executive - controls all programs/subprograms, (2) Translation - assembly language to machine language, (3) Execution control - actual test functions, (4) Utility - those routines for editing, debugging, etc., and (5) Maintenance - Self Test and CA/FI programs.

ID : 14  
SITE : wR-HLD  
SYSTEM : JTIDS  
SOFTWARE SUBSYSTEM : ASIT/OCP  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The Joint Tactical Information Distribution System (JTIDS) is an advanced system which provides Communications, Navigation, and Identification capabilities in an integrated form for application to military tactical and air defense operations. These capabilities are provided through the ability of the system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of pseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance, support, and intelligence information, weapons controllers, weapons systems, and decision-making commanders. JTIDS provides mobile surface and airborne force elements with a relative navigation capability within a common position reference grid and an intrinsic identification capability through the dissemination of crypto-secure position, velocity, and identity information concerning both friendly and hostile force elements.

The Adaptable Surface Interface Terminal (ASIT) equipment provides a transparent interface between existing ground command and control systems and the JTIDS network. The ASIT includes unique hardware and software along with a GFE Hughes Improved Terminal (HIT) with an IBM ML-1 Translator Processor (TP). The HIT will be referred to as the Class I Terminal. The ASIT converts the TADIL B Message Standard of the host platform/system into the Interim JTIDS Message Specification (IJMS) and vice versa.

CC : 21  
SITE : WR-ALL  
SYSTEM : JTIDS  
SOFTWARE SUBSYSTEM : E-TA AWACS/DOF  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The Joint Tactical Information Distribution System (JTIDS) is an advanced system which provides Communications, Navigation, and Identification capabilities in an integrated form for application to military tactical and air defense operations. These capabilities are provided through the ability of one system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of pseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance, support, and intelligence information, weapons controllers, weapons systems, and decision-making commanders. JTIDS provides mobile surface and airborne force elements with a relative navigation capability within a common position reference grid and an intrinsic identification capability through the dissemination of crypto-secure position, velocity, and identity information concerning both friendly and hostile force elements.

The Class 1 is a high powered terminal for use in the E-T and Adaptable Surface Interface Terminal (ASIT) Command, Control and Communications (C3) systems. It consists of both hardware and computer program software to participate in the JTIDS. The terminal provides the capability to transmit in assigned time slots within the network structure and to receive in all time slots not used for transmission. The Class 1 Terminal uses Interim JTIDS Message Specification (JMS) as its message format.

The E-T and ASIT version of the Class 1 terminal has no changes to the operational functions although it may be used as a final development test unit or as a test unit for testing and checkout purposes.

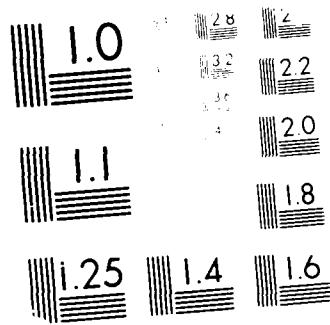
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SOFTWARE SUPPORTABILITY RISK ASSESSMENT IN OTAF  
DEPARTMENT OF DEFENSE AND DIA (U) BDM CORP ALBRIGHTWOOD MM  
D E PLENTY ET AL 07 OCT 85 BDM/R-85-0310-TR-001-1  
UNCLASSIFIED F29601-86-C-0035

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ID : E1  
SITE : WR-ALC  
SYSTEM : JTIDS  
SOFTWARE SUBSYSTEM : SP/USER  
SOFTWARE SUBSYSTEM TYPE : SIM  
DESCRIPTION:

The Joint Tactical Information Distribution System (JTIDS) is an advanced system which provides Communications, Navigation, and Identification capabilities in an integrated form for application to military tactical and air defense operations. These capabilities are provided through the ability of the system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of pseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance, support, and intelligence information, weapons controllers, weapons systems, and decision-making commanders. JTIDS provides mobile surface and airborne force elements with a relative navigation capability within a common position reference grid and an intrinsic identification capability through the dissemination of crypto-secure position, velocity, and identity information concerning both friendly and hostile force elements.

The Signal Processor User (SP/USER) Simulation Software is used to debug, test and exercise the Adaptable Surface Interface Terminal (ASIT)/E-3 Class I Operational Computer Programs (OCPs). It also simulates 3 users (ASIT and/or E-3) or one user and/or a JTIDS network.

DD : 22  
SITE : WR-ALC  
SYSTEM : JTIDS  
SOFTWARE SUBSYSTEM : SYS EXERCISER  
SOFTWARE SUBSYSTEM TYPE : SIM  
DESCRIPTION:

The Joint Tactical Information Distribution System (JTIDS) is an advanced system which provides Communications, Navigation, and Identification capabilities in an integrated form for application to military tactical and air defense operations. These capabilities are provided through the ability of the system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of pseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance, support, and intelligence information, weapons controllers, weapons systems, and decision-making commanders. JTIDS provides mobile surface and airborne force elements with a relative navigation capability within a common position reference grid and an intrinsic identification capability through the dissemination of crypto-secure position, velocity, and identity information concerning both friendly and hostile force elements.

The JTIDS System Exerciser (JSE) will be used to fully load the JTIDS net and create test scenarios for correcting problems. The JSE will enable real-time JTIDS tracks to be entered on the net independent from the surface subscriber source.

## THE BDM CORPORATION

ID : 23  
SITE : WR-ALC  
SYSTEM : PAVE TACK  
SOFTWARE SUBSYSTEM : AISF  
SOFTWARE SUBSYSTEM TYPE : SUP  
DESCRIPTION:

Pave Tack (AN/AVQ-26) is a 24 hour electro-optical target acquisition, laser/designator, and weapon delivery system for the United States Air Force. The system consists of a fuselage mounted pod and associated cockpit controls and display. It employs an Infrared Detecting Set which permits both day and night operation along with a relative adverse weather capability. In addition, it provides target location data to the aircraft weapons delivery digital computer to permit more accurate delivery of both conventional and guided ordnance. Pave Tack is currently configured for the USAF RF-4C, F-4E, and the F-111F aircraft.

The PAVE TACK Avionics Integration Support Facility (AISF) is an integrated set of hardware and software tools and avionics equipment used for the operational life support of the Pave Tack Operational Flight Program and Operational Test Program as used in the Pave Tack pod. The PTAISF provides capability to analyze impacts of Pave Tack user requests for changes, to modify the Operational Software, to verify and validate the modified software and to generate organizational maintenance level cassettes.

THE BDM CORPORATION

ID : 24  
SITE : WR-ALC  
SYSTEM : PAVE TACK  
SOFTWARE SUBSYSTEM : OFF  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

Pave Tack (AN/AVQ-26) is a 24 hour electro-optical target acquisition, laser/designator, and weapon delivery system for the United States Air Force. The system consists of a fuselage mounted pod and associated cockpit controls and display. It employs an Infrared Detecting Set which permits both day and night operation along with a relative adverse weather capability. In addition, it provides target location data to the aircraft weapons delivery digital computer to permit more accurate delivery of both conventional and guided ordnance. Pave Tack is currently configured for the USAF RF-4C, F-4E, and the F-111F aircraft.

The PAVE TACK Operational Flight Program (OFF) provides an interface function between the PAVE TACK pod, and the associated aircraft cockpit controls/display system and weapons delivery digital computer.

ID : 25  
SITE : 5M-ALC  
SYSTEM : F-111D  
SOFTWARE SUBSYSTEM : WNC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The navigation and weapon delivery system in the F-111 aircraft is an integrated avionics system. The heart of this system consists of two mission computers. One of these computers functions primarily as a weapons delivery computer and the other functions primarily as a general navigation computer. The Operational Flight Programs (OFF) loaded into these computers provides the navigation and weapon delivery computations and data required for automatic weapon delivery. Backup logic for most functions of each computer allows either computer to perform both navigation and weapon delivery functions in the event of a single computer failure. Some of the major modules in the OFFs are:

- (1) Navigation
- (2) Data Entry
- (3) Designation
- (4) Steering
- (5) Weapon Delivery
- (6) Air/Air Display

All F-111 OFFs written before 1965 were in IBM Assembly language and were hosted in IBM 4-PII computers. Current OFFs are written in Singer CPC-Ex Assembly Language and are hosted in Singer Weapons Navigation Computers (WNC). A rewrite of the FB-111A OFF into High Order Language (JOVIAL) is currently in progress.

ID : 26  
SITE : SM-ALC  
SYSTEM : F-111F  
SOFTWARE SUBSYSTEM : WNC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The navigation and weapon delivery system in the F-111 aircraft is an integrated avionics system. The heart of this system consists of two mission computers. One of these computers functions primarily as a weapons delivery computer and the other functions primarily as a general navigation computer. The Operational Flight Programs (OFP) loaded into these computers provides the navigation and weapon delivery computations and data required for automatic weapon delivery. Backup logic for most functions of each computer allows either computer to perform both navigation and weapon delivery functions in the event of a single computer failure. Some of the major modules in the OFPs are:

- (1) Navigation
- (2) Data Entry
- (3) Designation
- (4) Steering
- (5) Weapon Delivery
- (6) Air/Air Display

All F-111 OFPs written before 1985 were in IBM Assembly Language and were hosted in IBM 4-PI computers. Current OFPs are written in Singer CP2-EX Assembly Language and are hosted in Singer Weapons Navigation Computers (WNC). A rewrite of the FB-111A OFP into High Order Language (JOVIAL) is currently in progress.

ID : 27  
SITE : BM-ALC  
SYSTEM : FB-111A  
SOFTWARE SUBSYSTEM : WNC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The navigation and weapon delivery system in the F-111 aircraft is an integrated avionics system. The heart of this system consists of two mission computers. One of these computers functions primarily as a weapons delivery computer and the other functions primarily as a general navigation computer. The Operational Flight Programs (OFF) loaded into these computers provides the navigation and weapon delivery computations and data required for automatic weapon delivery. Backup logic for most functions of each computer allows either computer to perform both navigation and weapon delivery functions in the event of a single computer failure. Some of the major modules in the OFFs are:

- (1) Navigation
- (2) Data Entry
- (3) Designation
- (4) Steering
- (5) Weapon Delivery
- (6) Air/Air Display

All F-111 OFFs written before 1985 were in IBM Assembly Language and were hosted in IBM 4-PII computers. Current OFFs are written in Singer CFC-EX Assembly Language and are hosted in Singer Weapons Navigation Computers (WNC). A rewrite of the FB-111A OFF into High Order Language (JOVIAL) is currently in progress.

ID : 19  
SITE : CASTLE AFB  
SYSTEM : B-52  
SOFTWARE SUBSYSTEM : CPT  
SOFTWARE SUBSYSTEM TYPE : ATD  
DESCRIPTION:

The B-52 G/H Flight Simulator System Cockpit Trainer (B-52 CPT) is a uniquely tailored stand-alone software system which operates in real-time, receiving inputs from and transmitting outputs to the Flight Simulator and the Instructor Station.

ID : 22  
SITE : CASTLE AFB  
SYSTEM : B-52  
SOFTWARE SUBSYSTEM : WST  
SOFTWARE SUBSYSTEM TYPE : ATD  
DESCRIPTION:

The B-52 Weapons System Trainer (WST) will effectively support the training conducted at the Combat Crew Training Schools (CCTS) and Main Operating Bases (MOB) by providing capability in initial combat crew qualification, mission qualifications and continuation training for the maintenance of individual crew members and combined crew proficiency in the assigned tactical missions. Training that is directly transferable to the aircraft will be provided for B-52 Pilots, Co-pilots, Navigators, Electronic Warfare Officers, and Defensive Gunners.

These capabilities will be implemented through high fidelity simulation of flight and system characteristics and incorporation of instructional features designed to aid instructors in their task of instruction and performance evaluation.

Training in mission requirements from preflight to postflight may be accomplished and will include:

- (1) Mission planning
- (2) Transition training
- (3) Takeoff and landing training
- (4) Emergency procedure training
- (5) Instrument flight procedures and techniques
- (6) Aerial refueling operations
- (7) General navigation procedures
- (8) Celestial and pressure pattern navigation
- (9) Gravity weapon delivery
- (10) Special weapon delivery
- (11) Short Range Attack Missile/Air Launched Cruise Missile delivery
- (12) Terrain avoidance procedures and techniques
- (13) Threat evaluation and jamming
- (14) Tailgun Firing
- (15) Communication procedures
- (16) Crew coordination
- (17) Emergency war order training

ID : 70  
SITE : CASTLE AFB  
SYSTEM : KC-135  
SOFTWARE SUBSYSTEM : WST  
SOFTWARE SUBSYSTEM TYPE : ATD  
DESCRIPTION:

The KC-135 Weapons System Trainer (WST) will effectively support the training conducted at the combat Crew Training Schools and Main Operating Bases by providing capability in initial combat crew qualification, mission qualification, and continuation training for the maintenance of individual crew members and combined crew directly transferable to the aircraft. Will be provided for KC-135 pilots, copilots, and navigators. These capabilities will be implemented through high fidelity simulation of flight and system characteristics and incorporation of instructional features designed to aid instructors in their task of instruction and performance evaluation.

The KC-135 WST design is subdivided into four major subsystems: Flight, Navigator, Digital Radar Landmass, and Digital Image Generation (DIG) and includes the following:

- (1) Replica of the KC-135 Flight Station with positions for the pilot and copilot
- (2) Six-degree-of-freedom synergistic motion system for the Flight Station
- (3) Independent replica of the KC-135 Navigator Station with position for the Navigator
- (4) Modern CRT Instructor Stations
- (5) Digital Radar Landmass System
- (6) DIG Visual System
- (7) Fertkin-Elmer 8/32 computer complexes
- (8) Over-the-shoulder instructor positions on board each of the two stations

ID : 01  
SITE : CASTLE AFB  
SYSTEM : T-4 TRAINER  
SOFTWARE SUBSYSTEM : T-4 SIMULATOR  
SOFTWARE SUBSYSTEM TYPE : ATC  
DESCRIPTION:

The T-4 Simulator system simulates the Electronic Warfare (EW) portion of the E-8C. Basically all T-4 simulators consist of four major sections in addition to the power distribution panels. The instructor station section, system simulation section, and student station section make up the instructor equipment. The T4(VS) has four additional student stations and associated power panel.

ID : 32  
SITE : CO-ALC  
SYSTEM : F-10  
SOFTWARE SUBSYSTEM : FCC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The Fire Control Computer (FCC) performs five functions of primary importance to the F-10's weapon system: weapon delivery, energy management, MIL-STD-1580 multiplex bus control, navigation-related functions, and self test. The majority of the logical operations and mathematical computations needed to implement these five functions are carried out in the FCC. Additionally, the FCC interfaces with other subsystems such as the cockpit controls/displays, fuel measurement system, etc., through discrete inputs/outputs and analog inputs/outputs.

The software executed in the FCC is referred to as the FCC Operational Flight Program (OFP). The majority of the FCC OFP is written in the JOVIAL JCB-1 high order language and is mechanized in a modular structure. Minor elements of FCC logic are written in Magic Icc Assembly language. The FCC OFP is the only F-10 OFF written in high order language. The other six OFFs are written their applicable assembly languages. The FCC OFP is stored in magnetic core memory. The programming languages used were implemented prior to the USAF HCL policy. Cost and timing considerations preclude reprogramming these OFF's in HCL.

The FCC OFF provides logic and computations to implement and integrate fire control system modes and functions. The OFF consists of computer processing instructions which have been developed to satisfy allocated avionic requirements. Because of its central role in integrating F-10 sensors and equipment into the desired fire control system, the OFF is designated a configuration item and is managed in accordance with MIL-STD-482 and the configuration management plan (CPFF10).

The FCC OFP is a real-time program which coordinates sensor data, equipment data transfers over the serial digital multiplex data bus and schedules various processing activities to implement the fire control and navigation modes selected by the pilot.

ID : IT  
SITE : CO-4LC  
SYSTEM : F-16  
SOFTWARE SUBSYSTEM : HUD  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The new Head-Up Display (HUD) system element is a proven electro-optical design configured as a functional replacement to the HUD currently (1984) installed and operating in the F-16. The HUD is composed of three line replaceable units (LRUs): a display unit, an electronics unit, and a rate sensor unit. All of these are provided by Marconi-Elliott.

The HUD presents to the pilot visual flight and weapon delivery information as a forward real-world view while operating in a head-up position. The display provides a collimated image that is optically superimposed on the real-world view.

The display unit includes a combiner glass and mount capable of withstanding an air load of not less than 600 knots, a high brightness cathode ray tube that provides stroke-written symbology refreshed at a 50Hz rate, a night filter, a second manually depressible (0-210 mrd) reticle, and selectable symbol declutter control.

The electronics unit processes input data for symbol format displays. This repertoire includes symbology for 16 selectable operational modes. Also, the electronics unit provides the computation of the snapshot air-to-air gunnery solution and the back-up missile launch solution.

The rate sensor unit includes the measurement, processing and output of the air frame angular velocity components of roll, pitch, and yaw and normal acceleration as inputs for avionic systems computations.

ID : T4  
SITE : DC-ALC  
SYSTEM : F-16  
SOFTWARE SUBSYSTEM : OFT  
SOFTWARE SUBSYSTEM TYPE : ATD  
DESCRIPTION:

The Operational Flight Trainer (OFT) simulator cockpit is a replica of the F-16 aircraft cockpit. All instruments, fly-by-wire controls, avionics displays, and indicators are identical in appearance, color, feel and function to those of the F-16 aircraft design. All operating controls are monitored by the computational system, and resulting indications will be in response to the real-time software programs and avionic equipment. A mechanoreceptor cuing system comprised of a G-Seat, Anti-G Suit, and Seat Shaker is completely integrated into the cockpit to provide the needed motion indications. The fly-by-wire flight controls interact via signal conversion equipment (SCE) with the flight control system software model. An aural cue system will reproduce realistic aircraft sounds. The F-16 OFT is controlled and driven by a NORD-10/50 computer system. The computer system includes all required SCE, avionics multiplex bus interface, interfaces to visual, tactical, and instructional systems, simulator peripheral equipment, and operational software to control the simulation.

The Fire Control Computer, Stores Management System, Stores Control Panel, Central Interface Unit, Fire Control Navigation Panel, Heads-Up Display and Radar Electro-Optics are unmodified aircraft hardware, while the Remote Interface Units and the Fire Control Radar are simulated. Hardware and software produce air-to-air simulation and a representative air-to-ground simulation providing a pseudoground return combined with actual spatial modeling of surface targets. The remaining items of the aircraft avionics equipment, Central Update Computer, Target Identification Set Laser, Inertial Navigation System and Flight Control computer are respectively simulated in software to provide the proper stimuli to the physical avionic equipment.

ID : 35  
SITE : OO-ALC  
SYSTEM : F-16  
SOFTWARE SUBSYSTEM : FCR  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The F-16 Radar is a coherent pulse-doppler, multimode, digital fire control sensor designed to complement the air superiority and the strike roles of the F-16 multirole fighter. The radar is made up of six line replaceable units (LRUs): Antenna, transmitter, low-power radio frequency unit, digital signal processor, radar computer, and radar control panel.

The Fire Control Radar Operational Flight Program (FCR OFF) resides in the radar computer and interactively controls the functions of the other five LRUs. Communication with the other avionic computers is via the MUX bus. Additionally, the FCR interfaces with other subsystems through discrete inputs/outputs, video, analog input/output, and synchro input/output. The FCR OFF is written in assembly language and the OFF storage medium is erasable-programmable read-only memory.

The FCR OFF provides the processing necessary to implement F-16 radar system modes and functions, and to provide overall control of the radar hardware. The FCR OFF provides the required computations to perform the F-16 modes of operation. The FCR operation mode is determined by mode commands received from the Radar Control Panel, the Fire Control Computer, and the Stores Management Subsystem.

ID : 16  
SITE : 00-ALC  
SYSTEM : F-16  
SOFTWARE SUBSYSTEM : SMS  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The Stores Management System (SMS) provides for the monitoring, control, release, and jettisoning of stores on a selective and emergency basis. The SMS consists of three major components: Stores Control Panel (SCP), Central Interface Unit (CIU), and the Remote Interface Units. The SCP provides a continuous display of stores identification, location, quantity, percent status, and delivery mode.

Communication with other avionic computers is via the MILA Bus. Additionally, the SMS interfaces with other subsystems through discrete and analog inputs/outputs. The SMS Operational Flight Program (OFP) resides in the CIU in erasable-programmable read-only memory and is written in assembly language.

The SMS contains two microprocessors. The first microprocessor provides for the monitor, control and release of the loaded stores, and for jettison on a selective and emergency basis. The second microprocessor updates the SCP display and outputs data to the Bus. If either microprocessor or its associated memory or data busses malfunction, the other microprocessor will shut it down and will take over its functions. The SMS contains the provisions for a comprehensive self-test to minimize the amount of external testing required. The functions provided by the SMS include the following features:

- (1) Monitoring - display of store identification, location, quantity and present status.
- (2) Control - preparation for stores release through controls which allow pre-programming the SMS on the ground or during a mission.
- (3) Release - accomplishment of armed store release when requirements for release are satisfied.
- (4) Jettison - selective or emergency jettison of stores.
- (5) Mission Loading - acceptance of stores invention data into SMS memory via the SCP.

A number of pre-determined alternative programs for individual stores are present in the main memory. Each program contains a selection of all of the options (except Master Arm) necessary to ready a store for release, i.e., a weapon delivery option, an arming option, a release sequence option (such as step singles/pairs, ribbon singles/pairs, etc.) and, if applicable, multiple releases, the number of and separation between releases.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 37  
SITE : DO-ALC  
SYSTEM : F-4  
SOFTWARE SUBSYSTEM : MDT5  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 18  
SITE : DC-4LC  
SYSTEM : F-4E  
SOFTWARE SUBSYSTEM : AN/ARM-101  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : IS  
SITE : DC-MLC  
SYSTEM : F-4G  
SOFTWARE SUBSYSTEM : AN/ARM-101  
SOFTWARE SUBSYSTEM TYPE : CFF  
DESCRIPTION:

Not Available.

THE BOM CORPORATION

BOM/A-85-0510-TR

ID : 4G  
SITE : DC-4LC  
SYSTEM : F-4G  
SOFTWARE SUBSYSTEM : LRU-1 ADR  
SOFTWARE SUBSYSTEM TYPE : UFF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-TR

IC : 41  
SITE : GO-HLD  
SYSTEM : MINUTEMAN  
SOFTWARE SUB-SYSTEM : WING II 2015  
SOFTWARE SUB-SYSTEM TYPE : BDM  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BOM/A-85-0510-TR

ID : 42  
SITE : DODMLC  
SYSTEM : MINUTEMAN  
SOFTWARE SUBSYSTEM : WING V1.45-29  
SOFTWARE SUBSYSTEM TYPE : SIM  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-TR

CD : 4C  
SITE : OO-HLC  
SYSTEM : MINUTEMAN  
SOFTWARE SUBSYSTEM : WINGS/HB-26  
SOFTWARE SUBSYSTEM TYPE : SIM  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-72

CD : 44  
SITE : GO-HLC  
SYSTEM : MINUTEMAN II  
SOFTWARE SUBSYSTEM : 35AG/CAPS  
SOFTWARE SUBSYSTEM TYPE : SIM  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-72

IC : 48  
SITE : DO-48L  
SYSTEM : MINUTEMAN II  
SOFTWARE SUBSYSTEM : WING W/REG/RATE  
SOFTWARE SUBSYSTEM TYPE : GUF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BOM/A-85-0510-T2

CD : 46  
SITE : CO-HALC  
SYSTEM : MINUTEMAN II  
SOFTWARE SUBSYSTEM : WING VI REG RATE  
SOFTWARE SUBSYSTEM TYPE : SLF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BOM/A-85-0510-TP

ID : 47  
SITE : DC-ALC  
SYSTEM : RF-4C  
SOFTWARE SUBSYSTEM : AN/ARN-101  
SOFTWARE SUBSYSTEM TYPE : DPP  
DESCRIPTION:

Not Available.

IC : 48  
SITE : DC-48C  
SYSTEM : ALCM  
SOFTWARE SUBSYSTEM : LIT  
SOFTWARE SUBSYSTEM TYPE : ATE  
DESCRIPTION:

Level 1 Test (LIT) Software performs a field-depot level functional test of the AGM-65A Air Vehicle (ALCM). It supports Missile Sub Replacement Unit Fault Isolation, Inertial Platform Calibration and Air vehicle Certification after component replacement or missile upgrade. Major Test modules are:

- (1) ALCM/Electronic System Test Set Monitoring Circuits - verifies that monitoring circuits are within prescribed limits prior to applying power to the air vehicle.
- (2) ALCM Power - Verifies the air vehicle power tolerance.
- (3) ADCU Program Load and Verification - Loads and tests the load of the Air Vehicle Digital Unit (ADCU) test software.
- (4) Status and Fault Monitoring - Initializes monitoring of the Fault Isolation Record Table for the Level 1 performance test and verifies proper ADCU hardware operation.
- (5) INE Alignment and Navigation - Verifies execution of the INE Ground Alignment Sequence, the performance of the INE in the Navigation Mode and, that the effects of bias and scale factor of the vertical accelerometer are within tolerance.
- (6) Missle Radar Altimeter (MRA), Common Missile Radar Altimeter (CMRA) - Tests the radar altimeters.
- (7) Guided Missile Flight Controller Tests - Flight control off; Flight Control Power Superv Regulated; Movement of Engine thrust Controller; Gain and Frequency response of Flight Control Pitch/Roll Channels; movement of Air vehicle Elevons and; Aliveness of Telemetry Points.
- (8) Air Data Element - Verifies performance of the Pressure-Sensing Transducer and Electrical resistance Temperature Transmitter.
- (9) Inertial Reference Unit Torquing Test and INE Interface - Verifies the INE can torque the stable platform in roll and pitch and generate the correct phasing of the roll control coil channel error signal.
- (10) Flight Control Sensor Test - Verifies the FCS sensor electrical power, wheel speed detection circuitry and, flight control enable isolation buffer circuit.
- (11) Main DC Power, Rotary Switch and Warhead Commands - verifies main DC power bus integrity from the DC generator, separation switch performance and associated circuit integrity, and CHE/Warhead Safe/Armed/Pulsing commands.
- (12) Flight Control Pre-launch Test - verifies elevons are constrained when in a stowed position and that the INE Pre-launch Test can be successfully conducted.
- (13) INE Platform Calibration - calibrates certain inertial reference unit error parameters.

ID : 49  
SITE : DC-400  
SYSTEM : ALCM  
SOFTWARE SUBSYSTEM : LFT  
SOFTWARE SUBSYSTEM TYPE : ATE  
DESCRIPTION:

Loaded Pylon Test (LPT) Software is a field level functional test to verify operational status of loaded pylon and inertial platform calibration of all missiles, with or without warheads installed.

- (1) Continuity and Isolation - verifies that the correct patchboard has been installed, cooling air applied, pylon continuity and isolation and sets the pylon terminal address.
- (2) Power Application - applies 400 Hz. power to the pylon and to each missile. It also applies electronic power to each missile.
- (3) ADCU Program Load and verification - loads the mid vehicle Digital Computer Unit (ADCU) test software and verifies that this software has been correctly loaded into the ADCU of each missile.
- (4) INE Platform Calibration - This test is a calibration sequence to update certain inertial reference unit error parameters.
- (5) INE Ground Alignment and Navigation - verifies capability of the INE to execute properly the INE ground alignment sequence, the performance of the INE in the navigation mode and, that the effects of bias and scale factor of the vertical accelerometer have not exceeded acceptable limits.
- (6) Flight Control Pre-launch Test - verifies that the FCT pre-launch test can be successfully conducted.
- (7) Flight Control - Tests flight control functions including the roll and pitch channel integrators, cruise limit discretes, the DACR and DCCR.
- (8) IRU Torquing - verifies the Inertial Reference Unit (IRU) torquing about the INE platform roll and pitch axis.
- (9) Spin Down and Power Removal - used for gyro spin down and to remove power from the pylon and to verify the missile interface unit, MDU, and missile status.
- (10) Fast Data Block Screening Test - To detect a possible MDS modem and handshake intermittent failures.
- (11) Load Flight Software - Reapplies power after the test program data storage disc has been replaced with the flight software data storage disc and load and initialize the flight software into each missile.
- (12) Power Down - removes power from the pylon at the successful conclusion of the LPT.

ID : 80  
SITE : CC-ALC  
SYSTEM : ALCM  
SOFTWARE SUBSYSTEM : OFF  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The Air Launched Cruise Missile Operational Flight Program (ALCM OFF) provides all of the control logic required for captive alignment, free flight navigation and steering, terrain correlation and warhead control. The OFF executes a unique ALCM mission generated by the strategic mission planning system and loaded into the Air Vehicle Digital Computer Unit (ADCU) or the B-52 Offensive Avionics System (OAS) or B-1B Embedded Computer System. Specifically, the OFF is designed to do the following:

- (1) Provides commands to the Inertial Reference Unit to bring it into coarse alignment.
- (2) Monitors Inertial Navigation Element Built-In-Test circuits and tests avionics subsystems.
- (3) Computes Air Vehicle (A/V) position and velocity relative to earth.
- (4) Processes independent position measurement data to correct errors in the level channel navigation and inertial element alignment.
- (5) Compares the terrain altitude profile of the A/V flight path to prestored map areas.
- (6) Provides for safe transition from captive carry to free flight by activating subsystems to provide A/V flight control and propulsion.
- (7) Provides transitions of mission segments by processing A/V and waypoint position data.
- (8) Controls mode sequencing of the inertial platform for caging, power down, and nuclear recovery.
- (9) Uses A/V position data and mission data to compute vertical and lateral steering commands.
- (10) Computes the throttle command to maintain the A/V a prestored mach/time of arrival schedule.
- (11) Performs warhead arming by processing A/V position and target data to determine arrival at target and activates the warhead at the target via commands.
- (12) Processes temperature and pressure data to compute altitude, mach number, dynamic pressure and weight.
- (13) Restores itself to the pre-event configuration and restarts the system after a nuclear shutdown at the ADCU.
- (14) Commands the A/V to the proper conditions required for deployment of the recovery system for midair retrieval.
- (15) Provides the capability for the OFF to execute in free flight mode while the A/V is in captive carry.
- (16) Provides for the OFF to execute in the event of vacuum laboratory without initialization data from the carrier aircraft.

ID : 51  
SITE : OC-ALC  
SYSTEM : B-1B  
SOFTWARE SUBSYSTEM : CADC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The program requirement of the Central Air Data Computer (CADC) is to supply air data and stall warning data to the primary air data displays, flight control subsystem, avionics computer, and other aircraft subsystems. The basic functions supplied by CADC are as follows:

- (1) Provides altitude and rate of climb/descent calculation.
- (2) Provides mach/mach rate calculation.
- (3) Provides calibrated airspeed and acceleration calculation.
- (4) Provides true airspeed/acceleration, and temperature computation.
- (5) Supplies air data and stall warning data to the primary air data displays.
- (6) Provides operational performance data to Centralized Integrated Test System.

ID : 52  
SITE : OC-ALC  
SYSTEM : B-1B  
SOFTWARE SUBSYSTEM : CITS  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The B-1B Centralized Integrated Test System (CITS) provides for the collection and display of fault conditions in both avionics and non-avionics subsystems. Data identifying failures detected by the Central Computing System (CCS), as well as those detected by the TFACU and FACU, are consolidated into a single ACU (GNACU) and transmitted to the CITS dedicated computer. In performing those functions, CITS performs the following tasks:

- (1) Scheduling and timing control of functional modules.
- (2) System error handling.
- (3) Verification and selection of operator commanded CITS and Avionics Ground Readiness Tests.
- (4) Transient fault filtering.
- (5) CITS peripheral and computer self-test.
- (6) Scheduling and servicing of I/O operations.
- (7) Reading and recording of operator entered data.
- (8) Supplying responses to operator request.
- (9) Configuration of aircraft power per load management mode and test requirements.
- (10) Displays avionics messages.
- (11) Assembles messages for display, print and recording purposes.
- (12) Performs fault detection and isolation tests on the aircraft subsystems.

ID : 53  
SITE : OC-ALC  
SYSTEM : B-1B  
SOFTWARE SUBSYSTEM : EMUX  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The Electronic Multiplex System (EMUX) computer program provides for processing necessary to perform electrical control of aircraft subsystems via interface units called remote boxes. The program also provides the interface by which ground maintenance equipment can control and test aircraft subsystems. The EMUX program does not have a functional relationship to other computer programs. The primary functions which are implemented by this computer program are given below.

- (1) Control Box Data Transfer Function - This function provides for data transfer within the control box read-write memory. The processing includes transfer of remote box status data and loop-test data.
- (2) CGMI Function - This function provides for Central Integrated Test System Ground Maintenance Interface (CGMI) data processing. The function processes three words which contain aircraft subsystem control signals.
- (3) Control Box BIT Function - This function provides for performing the control box self-test functions. The processing consists of testing Boolean processor instructions, the accumulator, the FC Processor and the Caution Data Processor.
- (4) Load Management Function - This function provides processing necessary to determine the aircraft electrical load management mode. The function processes signals received from the various aircraft subsystems, the CGMI, and from the EMUX Mode Control function to generate a 4-bit code.
- (5) Caution Light Function - This function provides processing in support of the Caution Light Processor (CLP). The function generates a light flashing term value, and three other term values for the equations which the CLP evaluates in its processing.
- (6) Pre/Post FC Processing Function - This function provides for processing in support of the FC Processor. This function is performed before and after FC processing. The purpose of this function is to validate and configure FC data and to initiate the FC Processor. The Post FC processing part of this function provides processing for CGMI indication panel display.
- (7) EMUX Mode Control Function - This function provides processing for determining EMUX system mode of operation. There are two EMUX modes: flight mode and ground maintenance mode.
- (8) Aircraft Subsystem Control Functions - These functions provide processing for power control of aircraft subsystems. Though the subsystems are different, the processing performed in support of each is virtually the same.

ID : 54  
SITE : OC-ALC  
SYSTEM : B-1B  
SOFTWARE SUBSYSTEM : F/CGMS  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The Fuel/Center of Gravity Management System (F/CGMS) software is divided into two separate programs: Main Program and Signal Conditioning Program. Together, the Fuel/Center of Gravity Management Programs provide the processing necessary to measure and display fuel quantity, schedule fuel usage/transfer and control and display aircraft center of gravity. In addition, the F/CGMS processed inputs from the surface position sensors, the Central Air Data Computer (flight regime data), and from the Electronic Multiplex System (EMUX) (discrete inputs from the cockpit controls). It also provides outputs to EMUX (electrical control signals) and to Centralized Integrated Test System (BIT data). The basic functions performed by the F/CGMS software are as follows:

- (1) Provides an executive function which maintains control of the Operational Flight Programs (OFFPs).
- (2) Provides an input/output function.
- (3) Provides a fuel quantity correction function which corrects the fuel quantity valve for variations in fuel reference signal, fuel contamination, and fuel density.
- (4) A center of gravity function determines the aircraft Center of Gravity and its limits.
- (5) A fuel usage scheduling function provides for fuel pump and fuel valve controls.
- (6) A test function performs CPU memory, signal conditioners, and F/CGMS system indicators test.
- (7) A ground test function provides ground test routines to perform aircraft ground checkout.
- (8) Development and test software will apply to each module.

THE BDM CORPORATION

ID : 55  
SITE : OC-ALC  
SYSTEM : B-1B  
SOFTWARE SUBSYSTEM : INS  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The Inertial Navigation System (INS) provides the navigational data utilized by the Central Computing System (CCS) and Offensive Radar System. The basic functions of the INS operational software are as follows:

- (1) Real-Time Executive
- (2) Inertial Measurement Unit (IMU) Alignment/Preflight Calibrator
  - (3) Provides Navigational calculation information to the CCS
- (4) Provides steering outputs to the CCS
- (5) Provides IMU Torquing
- (6) Performs Built In Test
- (7) Processes Mux Messages
- (8) Provides various utility programs for support
- (9) Provides for System Mode Control

## THE BDM CORPORATION

ID : 56  
SITE : DC-ALC  
SYSTEM : B-1B  
SOFTWARE SUBSYSTEM : ORS  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The B-1B Offensive Radar Subsystem (ORS) operational flight program is used to control the Programmable Signal Processor (PSP) and the Radar Display Electronics Unit (RDEU) as necessary to accomplish radar functions. The PSP is composed of two functional computing elements, the Radar Computer (RC) and the Array Processor (AF). The RC in general performs the radar control and external interface functions while the AF performs the radar detection data processing. The RDEU contains a general purpose computer which interfaces with the PSP and controls the display of information on the Radar Display Unit. Two basic modes of operation are below:

- (1) Multi-Mode Radar Mode - This inclusive mode consists of many modes performing the following functions:
  - (a) Provides quiet mode for non-radiating periods between active modes
  - (b) Provides high resolution ground map to provide SAR surface mapping for navigation and target location
  - (c) Provides accurate measurement of navigation position errors in Position Update mode
  - (d) Provides low altitude terrain evaluation profiles in Terrain Following mode
  - (e) Provides an azimuth vs. range display of terrain above a specified altitude in Terrain Avoidance mode
  - (f) Provides surface mapping for navigation and target location in Real Beam Ground Map mode
  - (g) Provides navigation position update with respect to the known locations of fixed ground beacons in Ground Map Beacon mode
  - (h) Provides azimuth vs. range display of rainfall in front of aircraft in Weather mode
  - (i) Provides interrogation/tracking of airborne beacon for manual rendezvous with tankers in Rendezvous Beacon mode
  - (j) Provides skin-return tracking of other aircraft
  - (k) Ground Moving Target Identification/Tracking mode detects/tracks fast moving ground targets in Rendezvous mode
  - (l) Measures aircraft ground velocity for navigation alignment in Velocity Update mode
  - (m) Measures aircraft altitude above terrain beyond range of the altimeters in High Altitude Calibrate mode
  - (n) Provides ranging data to a designated ground location in Air to Ground Ranging mode
- (2) Radar Terrain Sensor Mode - The following modes, described in the multi-mode radar mode section, are included: quiet, terrain following, terrain avoidance, real beam ground map, position update, and high altitude calibrate.

THE BDM CORPORATION

BDM/A-85-0510-TD

IC : E7  
SITE : DC-HALO  
SYSTEM : E-50  
SOFTWARE SUBSYSTEM : BNST  
SOFTWARE SUBSYSTEM TYPE : AFD  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BCM/A-85-0510-T2

CC : 56  
SITE : DC-HALC  
SYSTEM : B-52  
SOFTWARE SUBSYSTEM : FTSS  
SOFTWARE SUBSYSTEM TYPE : ACF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

ID : 54  
SITE : DC-HQD  
SYSTEM : B-52  
SOFTWARE SUBSYSTEM : MD-1 EXEC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BOM/A-85-0510-72

IC : 80  
SITE : DC-442  
SYSTEM : B-82  
SOFTWARE SUBSYSTEM : MC-2 EXEC  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

Not Available.

IE : SI  
SITE : DC-AHDS  
SYSTEM : E-3A  
SOFTWARE SUBSYSTEM : INS  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (C3) center in a modified Boeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Inertial Navigation System (INS) Computer Program provides navigation information to the Airborne Operational Computer Program through the Control Power Supply. It provides for automatic navigation and guidance through the computations of inertial-sensed accelerations augmented by inputs and guidance from external avionics equipment. It includes the following capabilities:

- (1) Determine angular orientation and horizontal velocity of the aircraft.
- (2) Determine ground speed and drift angle.
- (3) Determine geographic position.
- (4) Generate and supply a steering signal for control of flight.
- (5) Generate and supply attitude reference signal to flight instruments.
- (6) Determine the north and east velocity of the aircraft.

ID : 52  
SITE : GC-AAC  
SYSTEM : E-3A  
SOFTWARE SUBSYSTEM : OMEGA  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (C3) center in a modified Boeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Omega Navigation Computer Program provides corrections to the navigation data in the Inertial Navigation Equipment Computer Program. It provides automatic navigation and guidance through the computations of Omega transmission signals and Doppler velocity measurements augmented by inputs from external avionics equipment. The Omega Navigation Computer program combines the inputs from the Inertial, Omega, and Doppler sensors in a Kalman filter, which then provides corrections to the inertial equipment.

ID : 57  
SITE : CC-HLC  
SYSTEM : E-3A  
SOFTWARE SUBSYSTEM : SMCF  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (CC) center in a modified Boeing 707 aircraft. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs including programs for Automatic Test Equipment (ATE).

The System Maintenance Computer Program (SMCF) operates in real time with, and under the direct control of, the Airborne Operational Computer Program (AOCP) Executive. SMCF execution is interleaved with execution of AOCP tasks and the In-Flight Performance Program.

The SMCF is made up of the following four functions:

- (1) Monitor And Test Subsystem Control - Provides fault detection and isolation for the displays of the Identification Functional Group and On-Board Test Monitor and Maintenance Functional Group (OBTM&MFG) through control and interrogation of OBTM&MFG test points. Test sequences and failure criteria are specified by performance monitoring and fault isolation trees.
- (2) Confidence Testing - Provides for detection of Line 2 Power Supply (LPS) and avionics interface faults through control and monitoring of wrap around tests and LPS subsystem status report.
- (3) Display Dialog - Provides for isolation of display failures by use of a man-machine dialog consisting of display test patterns, messages, and operator replies via switch actions.
- (4) Central Maintenance - Processes equipment status reports received from other SMCF functions, the AOCP Executive, and the computer operator. It maintains equipment status tables for use in device allocation and system reconfiguration; initiates display messages and line printer output; notifying operator of incidents and providing replace media instructions; and records maintenance history data or management data for data reduction on the ground.

ID : 64  
SITE : DC-AFC  
SYSTEM : E-3A  
SOFTWARE SUBSYSTEM : SRCP  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (C3) center in a modified Boeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Surveillance Radar Computer Program (SRCP) operates on the E-3 Radar Data Correlator (RDC) and sends preprocessed data to the Airborne Operational Computer Program. The RDC consists of a dual processor with separate core program memory and MCS data memory, a special hard-wired processor for pulse doppler range resolution, and an I/O unit for communicating with the radar subsystems and the Control Power Supply. The SRCP is organized into a main program, normally resident in the RDC, and a fault isolation test library, which resides off-line on magnetic tape. The SRCP is divided into three functional areas: Data Processing And Control (DPAC), Fault Detection (FD), and Fault Isolation Test (FIT).

(1) The DPAC software provides the specific radar functions of input/output (I/O) control and data sequencing, data memory allocation management, mode control, beam stabilization, main beam clutter tracking, range resolution, correlation of radar returns over multiple modulation periods, data processing for pulse doppler, target formatting, and a maritime surveillance capability.

(2) The FD software provides continuous monitoring of various GO/NO-GO fault indications of the radar. Interleaved tests are performed to diagnose faults in the RDC or in the communications links with other radar subsystems. Dedicated time tests and manually selectable tests provide detailed diagnosis of radar units. The FD software controls execution of all tests during turn-on and normal operation. If parameters or test results require it, the FD software controls reconfiguration of the radar by switching in redundant units.

(3) The FIT software consists of detailed tests to align radar faults to replaceable units in major radar subsystem elements. These tests normally reside off-line, and, when requested manually or automatically, are loaded into the RDC.

ID : 68  
SITE : DC-4LC  
SYSTEM : E-3A  
SOFTWARE SUBSYSTEM : SRGSCF  
SOFTWARE SUBSYSTEM TYPE : SUP  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (C3) center in a modified Boeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs (including programs for Automatic Test Equipment (ATE)).

The Surveillance Radar Ground Support Computer Program (SRGSCF) provides the support software needed to generate, maintain, and test the SRCP and SRMCF. It consists of the following functional components:

- (1) Program Generation Package (PGP) - Provides for production of the SRCP tapes and maintenance of the radar program files. Includes the RDC assembler and loader.
- (2) Radar Data Generator (RDG) - Generates realistic radar target and ECM detection data from a scenario input for exercising the SRCP.
- (3) RDC Functional Simulator (RDCFS) - Simulates the RDC processor and data transfer for active and passive I/O for testing the SRCP on the IBM 370/168.
- (4) Digital Land Mass Blanker Map Tape Generation Program (DLMB MTGP) - Generates and/or modifies Land Mass Blanking maps on magnetic tapes that are read into the DLMB unit.

PGP, RDG, and RDCFS run on the IBM 370/168 and are written mainly in IBM 370 Assembly language. DLMB MTGP is written mainly in FORTRAN.

THE BDM CORPORATION

ID : 80  
SITE : 00-420  
SYSTEM : BLCM  
SOFTWARE SUBSYSTEM : SFS  
SOFTWARE SUBSYSTEM TYPE : SFS  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-18

ID : 97  
SITE : DC-HLC  
SYSTEM : GLEM  
SOFTWARE SUBSYSTEM : M-BTD  
SOFTWARE SUBSYSTEM TYPE : BLF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 28  
SITE : DC-4LC  
SYSTEM : BLOM  
SOFTWARE SUBSYSTEM : MEF  
SOFTWARE SUBSYSTEM TYPE : GOF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-TR

IC : 59  
SITE : DC-HLD  
SYSTEM : GLCM  
SOFTWARE SUBSYSTEM : OFF  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-0510-12

ID : 70  
SITE : DC-HLD  
SYSTEM : GLCM  
SOFTWARE SUBSYSTEM : WCS  
SOFTWARE SUBSYSTEM TYPE : DCE  
DESCRIPTION:

Not Available.

THE BDM CORPORATION

BDM/A-85-051G-TR

10 : 71  
SIZE : 1024K  
SYSTEM : SRAM  
SOFTWARE SUBSYSTEM : OFF  
SOFTWARE SUBSYSTEM TYPE : OFF  
DESCRIPTION:

Not Available.

IC : 72  
SITE : TINKER AFB  
SYSTEM : E-3A  
SOFTWARE SUBSYSTEM : ACCF  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (C3) center in a modified Boeing 707 airframe. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs including programs for Automatic Test Equipment (ATE).

The Airborne Operational Computer Program (ACCF) is a compilation of real-time programs that execute on the aircraft during an airborne mission. In general, ACCF processes navigation and radar data, performs calculations, and formats information for display on the E-3 consoles. It is written in COBOL Language and Assembly Language and operates on the DAPG 4P, DC-1 or CC-1 Data Processing System. Specifically, ACCF processes data received from all avionics subsystems and from unrecorded magnetic tapes to perform the intended mission. It assists the on-board personnel in the following areas:

1. Detect, track, and identify air traffic.
2. Commit and control weapon resources.
3. Communicate with external interfaces and relay communication messages.
4. Display and process data for on-board operations.
5. Perform on-board training.
6. Record data for analysis in the Computer Program Ground Support Center.

ACCF functions include the CHFC Executive, Surveillance, Weapons Control, Communications, Displays, Switch Actions, Internal Simulation, Battle Staff, and the System Maintenance Computer Program execution. These programs are the very core of the E-3 system.

ID : 77  
SITE : TINKER AFB  
SYSTEM : E-3A  
SOFTWARE SUBSYSTEM : UTILITIES  
SOFTWARE SUBSYSTEM TYPE : SUF  
DESCRIPTION:

The E-3 Airborne Warning and Control System (AWACS) is a high-capacity radar station and command, control, and communication (C2) center in a modified Boeing 707 airplane. The system includes its associated ground support facilities and equipment. The airborne equipment is composed of five integrated computers and their software. The ground support equipment includes software-driven simulators, trainers, and test equipment, and computer hardware, software, and facilities to support the E-3 computer resources. Each version of E-3 software consists of over 700 individual computer programs including programs for Automatic Test Equipment (ATE).

The Utility Computer Program (UCP) provides data for the Airborne Operational Computer Program (AOCP) in support mission deployment. It supports generation, test, and maintenance of E-3 computer programs and data bases. The UCP includes the following functions.

Generates binary map files for display on Situation Display Consoles from alphanumeric inputs consisting of latitude and longitude and display control information. Provides the capability of presetting values in the AOCP data base. Generates or updates E-3 data bases from input adaptation tables and JCS/AFI COMFOCL symbol definition information for use in the E-3 operating environment. Assembles programs written in AFM Data Processing System Assembly Language. Includes seventeen programs and several support routines supporting three functions: control, compilation, and COMFOCL assembly. Generates program and adaptation tapes for use on the AFM CC-1 or CC-2 data processor by retrieving program and data elements from secondary storage volumes and transferring them to magnetic tape in a machine-useable format. Receives as input the magnetic tape containing information produced by the Unit Test Controller Function. Selected portions are formatted and directed to printer for output. Includes a machine instruction level simulator that provides a capability to test AFM CC-1 or CC-2 computer program units on the UFGSC computer. Includes a utility subprogram library of commonly used mathematical, data conversion, and data manipulation subroutines.

IC : 74  
SITE : LANGLEY  
SYSTEM : JTIDS  
SOFTWARE SUBSYSTEM : ASIT-TFOCP  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The Joint Tactical Information Distribution System (JTIDS) is an advanced system which provides communications, navigation, and identification capabilities in an integrated form for application to military tactical and air defense operations. These capabilities are provided through the ability of the system to distribute information quickly and encrypted to provide security and reliability in hostile environments. Security and jamming resistance are obtained through the use of pseudorandom signal processing techniques. The system provides a capability to interconnect scattered sources of surveillance, support, and intelligence information, weapons controllers, weapons systems, and decision-making commanders. JTIDS provides mobile surface and airborne force elements with a relative navigation capability within a common position reference grid and an intrinsic identification capability through the dissemination of encrypted position, velocity, and identity information concerning both friendly and hostile force elements.

The Translator Processor Operational Computer Program (TPOCP) software is one part of the Adaptable Surface Interface Terminal (ASIT) system (the other is the ASIT Class I Functional and its software, the Communications Processor Operational Computer Program (CPOCP)). The TPOCP provides a transparent translation of tactical air control messages traffic between the interim JTIDS Message Specification format and the Tactical Air Control System Tadlit-B format, for one to five subscribers. The subscribers are Air Force Army ground-based control and reporting centers or Message Processing Centers.

The TPOCP performs several basic functions in providing the transparent interface for the ASIT. The primary function of the TPOCP is to translate JTIDS messages to Surface surveillance SS messages and to translate SS messages to JTIDS messages. Other functions include the maintenance of the data base of all active tracks, the implementation of the Joint Tactical Air Operations transmission rules, message acknowledgement, and the control of the transmission and reception of messages. The TPOCP also provides for:

1. Statistics gathering and fault detection over the system for trouble monitoring
2. Redundant or operational statistics and fault tolerance information data collection
3. The ability to receive bidirectional communications from the TADLIT-B system (ASIT) via CPOCP interface (TPOCP).

ID : 75  
SITE : Langley  
SYSTEM : STRTS  
SOFTWARE SUBSYSTEM : STRTS  
SOFTWARE SUBSYSTEM TYPE : ATO  
DESCRIPTION:

The Simulator Tactical Radar Training System (STRTS) is a simulator system to support training requirements of the 407L operational system. STRTS will provide an improved system for training and exercising the Tactical Air Control System, Control and Reporting Center/Control and Reporting Post operations personnel. The STRTS consists of digital computers with peripheral equipment; a video generator; positions for one simulation supervisor, one simulation supervisor technician, six pilot simulators, and a computer operator; automatic inter-system data link equipment; voice communications; and the operational, diagnostic and support software to control system functions. The STRTS can be housed in a van (for portability) or a fixed facility.

ID : T-5  
SITE : Langley  
SYSTEM : TACCS  
SOFTWARE SUBSYSTEM : CAFMS  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The Tactical Air Control Center (TACC) is the operations center of the Tactical Air Control System (TACS). The mission of the TACC is to prepare, disseminate, and monitor the execution of coordinated orders for the employment of all the forces assigned, attached, or otherwise made available to the Air Force Component Commander. The manual TACCS procedures are inadequate to handle the increasing complexity and amount of operational information. Computer Assisted Force Management System (CAFMS) provides an automated assist to the TACCS through information storage and retrieval along with secure digital communications. A minicomputer located at the TACC will support up to 12 local terminals and 13 remote terminals. The remote terminals have limited stand-alone capabilities and will be located at the Wing Operations Center (WOC), Air Support Operations Center (ASOC), or Control and Reporting Center (CRC). Each terminal consists of a display unit, a keyboard, and a table top printer. Remote terminals also have a floppy disk.

The CAFMS software provides automated assistance to two of the TACC functions, Combat Plans and Combat Operations. Combat Plans will use CAFMS to construct, review, and disseminate the Air Tasking Order, generate operational mission schedules, and monitor TACC resources. Combat Operations will use CAFMS to follow mission progress, generate recaps and reports, and monitor TACC resources. In addition, CAFMS will expedite information transfer between the TACC and lower echelon elements including the WOC, ASOC, and the CRC.

ID : 77  
SITE : LANGLEY  
SYSTEM : TIFI  
SOFTWARE SUBSYSTEM : DC/SR  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The Tactical Information Processing and Interpretation (TIFI) System is comprised of several segments. These currently include the Display and Control Storage and Retrieval Segment; the Imagery Interpretation (II) Segment; and the Manual Radar Reconnaissance Exploitation System (MRRS) Segment.

The DC/SR system is a mobile, sheltered, computer-based Data Management System used by the intelligence division as their master intelligence data base. The DC/SR software provides the capability for a self-contained (e.g., noncommercial) communications capability to transmit and receive message (e.g., (1) traffic via teletype, digital data link, hard copy, and voice communication lines. This message flow is essential to the Tactical Air Control Center (TACC), since it is the only means by which the combat plans division can receive and transmit intelligence information.

The DC/SR segment provides automated assistance to the operational intelligence, collections and targeting functions associated with the TACC pre-planned air tasking order nomination and production cycle. Digital installation and order of battle data base files are maintained and updated within the DC/SR. A capability for weaponeering, and automated report-situation generation and dissemination is also available in the segment. Its communication subsystem provides automated assistance for the receipt, processing and transmission of messages, and allows for dedicated digital data link interfaces with the II segments and the AUTODIN network.

ID : 73  
SITE : LANGLEY  
SYSTEM : TIFI  
SOFTWARE SUBSYSTEM : II/MARRES/TEREC  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The Tactical Information Processing and Interpretation (TIFI) System is comprised of several segments. These currently include the Display and Control/Storage and Retrieval (DCSR) Segment; the Imagery Interpretation (II) Segment; and the Manual Radar Reconnaissance Exploitation System (MARRES) Segment.

The II software provides computer assisted interpretation and exploitation of photo imagery reconnaissance data. The mission derived from early 1964 Vietnam requirements for photo interpretation. In 1981 the Tactical Electronic Reconnaissance (TEREC) capability was added to provide Electronic Intelligence (ELINT) at near real-time information processing capability. The MARRES software provides computer assisted interpretation and exploitation of Side Looking Radar Imagery.

The II Segment employs automated light stations and associated ADF equipment and computer programs to expedite the processing and formatting of photographic imagery and ELINT data into meaningful intelligence report. These automated facilities better equip the analyst to fully exploit the increased volume rate and sophistication of present-day imagery by providing the following capabilities: rapid scanning, magnified viewing, and accurate computer aided mensuration of photographic imagery; rapid initial interpretation of photographic imagery to produce intelligence information which is time critical; detailed interpretation and analysis of photographic imagery including collation with data base information collected by other means; and computer assistance in exploiting the data from MILSTD-152 Code clocks.

The MARRES Segment is a basic shelter integrated with hardware and software designed to be employed by the USAF for the purpose of exploiting radar imagery.

The MARRES has been developed to provide the automated equipment, techniques, and procedures required to significantly increase the intelligence capabilities of the echelons where it will be employed. The MARRES is designed to accomplish the following basic functions: pre-mission planning and preparation of collated data; analysis; interpretation of AN/UPD-3 (formerly AN/UPD-4) type radar imagery; generation of intel intelligence reports; and dissemination of analyzed data.

ID : 79  
SITE : Langley  
SYSTEM : 407L  
SOFTWARE SUBSYSTEM : HUGHES UTIL  
SOFTWARE SUBSYSTEM TYPE : SUP  
DESCRIPTION:

The 407L Computer Programs were designed in a modular fashion to facilitate program maintenance and modifications; planned growth items may be added with minimum impact. The state-of-the-art has been advanced in diagnostic programs, tracking logic, on-line site adaptation, and site registration. The net effect is a software package which is easily maintained and capable of modular expansion to meet future contingencies. The 407L programs include operational and recording, utility support for simulation and data reduction, system utility support, and diagnostic troubleshooting.

The Simulation and Data Reduction Programs operate off-line in support of training, system exercise, and program maintenance. The Simulation Programs generate magnetic tape inputs of simulated data which are processed on-line by the Operational Programs. Target reports, switch actions, data link inputs, and tabular data may be simulated. The Data Reduction Programs process data recorded on magnetic tape by the Operational Programs and provide hardcopy output for post-mission debriefing or program checkout.

The System Utility Programs are used for computer program production and maintenance. The Assembler converts symbolic computer instructions into machine code and provides a library of mathematical routines. The remainder of the Utility Programs provide program debugging aids.

The Diagnostic Programs provide on-line System Status monitoring and off-line equipment troubleshooting. The equipment design and design of the diagnostic routines were conducted in concert to provide fault detection capability with isolation to a functional card group of from one to ten cards.

THE BDM CORPORATION

BDM/A-85-0510-TR

ID : 30  
SITE : Langley  
SYSTEM : 407L  
SOFTWARE SUBSYSTEM : IBM UTIL  
SOFTWARE SUBSYSTEM TYPE : SLP  
DESCRIPTION:

The IBM UTIL is essentially a conversion and enhancement of HUGHES UTIL software to an IBM computer environment. It has a functional description similar to the HUGHES UTIL software.

ID : 31  
SITE : Langley  
SYSTEM : 407L  
SOFTWARE SUBSYSTEM : IORP/IMPF  
SOFTWARE SUBSYSTEM TYPE : C-E  
DESCRIPTION:

The 407L Computer Programs were designed in a modular fashion to facilitate program maintenance and modifications; planned growth items may be added with minimum impact. The state-of-the-art has been advanced in diagnostic programs, tracking logic, on-line site adaptation, and site redistribution. The net effect is a software package which is easily maintained and capable of modular expansion to meet future contingencies. The 407L programs include operational and recording, utility support for simulation and data reduction, system utility support, and diagnostic troubleshooting.

The Operational and Recording Programs accomplish real-time command and control processing in support of the prime JFC/JFF AC&W mission. The two primary functions controlled by the operational programs are surveillance and weapons control. The Surveillance Programs accomplish automatic and rate-aided-manual tracking, identification processing, crosstail and automatic intersite radar alignment. Planned growth features in the surveillance area include automatic shift tracking, and flight plan processing. The Weapons Control Programs provide automatic guidance computations for offensive and defensive mission (voice) control. The weapons programs accommodate three tactic and profile options structured for tactical air operations. Planned growth features in the weapons area are trial intercept computations for weapons assignment, fuel/weapons status monitoring, and ground-to-chain data link. The General Data Programs accomplish such program control and input/output functions as executive processing, display generation, switch action processing, data recording, and peripheral equipment input/output processing. The input/output processing for the automatic data link and radar processor also falls under general data processing.

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