

MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

AD-A156 540

①
Etk

ESTIMATION OF F-15 PEACETIME
MAINTENANCE MANPOWER REQUIREMENTS
USING THE LOGISTICS COMPOSITE MODEL

THESIS

GOR/SM/76D-5

George DeGiovanni
Captain USAF

Donald M. Douglas
Major USAF

DTIC FILE COPY

A. F. IP
Wright
MAR 1977

DTIC
SELECTED
JUL 1 1 1985
S E D

This document has been approved
for public release and sale; its
distribution is unlimited.

85 7 01 176

Abstract

Estimates of the maintenance manpower requirements of a planned F-15 Tactical Fighter Training Wing operating in a peacetime environment were requested by Tactical Air Command. The Logistics Composite Model (LCOM), a computer simulation, was used in conjunction with the Moody Regression and Moody Manpower programs to estimate these requirements. A more efficient method, using statistical tests of hypothesis, was developed for determining steady state in the simulation model. Using this method, steady state conditions were found to exist, in most cases, at the end of the first simulated day. An estimate was made of the autocorrelation present in each set of simulation output data. Then, correcting for this autocorrelation, statistical confidence intervals were constructed for the manpower estimates. By simulating at various levels of flying activity and with various constraints on resource availability, manpower requirements were found to be relatively insensitive to these constraints at low sortie rates and more sensitive at higher sortie rates. The authors suggest that the construction of statistical confidence intervals and the methodology developed in this study for determining steady state should be given serious consideration in future LCOM manpower studies. *... and keywords include: no 1473*

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER GOR/SM/76D-5	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) ESTIMATION OF F-15 PEACETIME MAINTENANCE MANPOWER REQUIREMENTS USING THE LOGISTICS COMPOSITE MODEL		5. TYPE OF REPORT & PERIOD COVERED
		6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(s) George DeGiovanni, Capt, USAF Donald M. Douglas, Maj, USAF		8. CONTRACT OR GRANT NUMBER(s)
9. PERFORMING ORGANIZATION NAME AND ADDRESS Air Force Institute of Technology (AFIT/EN) Wright-Patterson AFB, Ohio 45433		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
11. CONTROLLING OFFICE NAME AND ADDRESS Air Force Institute of Technology (AFIT/EN) Wright-Patterson AFB, Ohio 45433		12. REPORT DATE December 1976
		13. NUMBER OF PAGES 196
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		15. SECURITY CLASS. (of this report) UNCLASSIFIED
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) APPROVED FOR PUBLIC RELEASE AFR 190-17.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES Approved for public release; IQW AFR 190-17 <i>J. F. Guess</i> JERAL F. GUESS, Captain, USAF Director of Information		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Model Manpower Estimate Simulation Autocorrelation.		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Estimates of the maintenance manpower requirements of a planned F-15 Tactical Fighter Training Wing operating in a peacetime environment were requested by Tactical Air Command. The Logistics Composite Model (LCOM), a computer simulation, was used in conjunction with the Moody Regression and Moody Manpower programs to estimate these requirements. A more efficient method, using statistical tests of hypothesis, was developed for determining steady state in the simulation model. Using this method, steady state conditions were found to exist, in most cases, at the end of the first simulated day. An estimate was		

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

made of the autocorrelation present in each set of simulation output data. Then, correcting for this autocorrelation, statistical confidence intervals were constructed for the manpower estimates. By simulating at various levels of flying activity and with various constraints on resource availability, manpower requirements were found to be relatively insensitive to these constraints at low sortie rates and more sensitive at higher sortie rates. The authors suggest that the construction of statistical confidence intervals and the methodology developed in this study for determining steady state should be given serious consideration in future LCOM manpower studies.

04 M1410XXZ

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

ESTIMATION OF F-15 PEACETIME
MAINTENANCE MANPOWER REQUIREMENTS
USING THE LOGISTICS COMPOSITE MODEL

THESIS

Presented to the Faculty of the School of Engineering
of the Air Force Institute of Technology
Air University
in Partial Fulfillment of the
Requirements for the Degree of
Master of Science

by

George DeGovanni, B.S.
Captain USAF

Donald M. Douglas, B.S.
Major USAF

Graduate Operations Research

December 1976

Accession For	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By _____	
Distribution/ _____	
Availability Codes	
Dist	Avail and/or Special
A-1	



Preface

This thesis represents the first detailed Logistics Composite Model (LCOM) analysis of United States Air Force (USAF) peacetime flying operations. In the study, we have attempted to shed light on the statistical and sensitivity inferences of the manpower estimation process. Hopefully, our efforts will allow the LCOM community to achieve greater statistical accuracy with LCOM.

We would like to thank our advisor, Colonel Ronald A. Luhks, and reader, Lieutenant Colonel Jon R. Hobbs, for their guidance during these past six months. We especially thank Lieutenant James R. Lowell for his ceaseless efforts and expert technical advice throughout our thesis effort. We also would like to thank Lieutenant Colonel Donald C. Tetmeyer, William D. Moody, and Wayne Jansen for their help with LCOM procedures. Finally, we would like to thank our wives, Mary and Bobbi, without whose help and encouragement this thesis would not have been possible.

George DeGiovanni

Donald M. Douglas

Contents

	<u>Page</u>
Preface.	ii
List of Figures.	v
List of Tables	vii
Symbols and Abbreviations.	viii
Abstract	xi
I. Introduction	1
Background.	1
Thesis Objectives	2
Thesis Scope.	3
Overview.	3
II. LCOM Manpower Estimation	5
Logistics Composite Model	6
Pre-Processor Program.	9
Main Program	14
Post-Processor Program	16
Moody Regression Program.	16
Moody Manpower Program.	16
Summary	17
III. F-15 TFTW Data Base.	19
Maintenance Data Base	19
Maintenance Organization Structure	19
Scheduled and Unscheduled Maintenance.	24
Maintenance Repair/Service Times	25
Weapon System Components	25
Component Failure Parameters	25
Operations Data Base.	26
Flying Schedule.	27
Maintenance Schedule	28
Resource Constraints.	28
Summary	29
IV. Methodology.	30
Sequence of Simulation Runs	30
Determination of Manpower Requirements.	30
Model Validation.	33
Steady State.	35
Some Criteria for Steady State.	37
Student's T Test	37
Mann-Whitney U Test.	38
Use of the U Test in Determining Steady State	38
The Runs Test.	40
Type I and Type II Errors.	42

Contents

	<u>Page</u>
Confidence Intervals	42
Autocorrelation	43
Correcting for Autocorrelation	44
Sensitivity Analysis	45
Developing a Manning Document	45
Summary	46
V. Analysis and Results	47
Analysis	47
Steady State	47
Autocorrelation	56
Results	58
Direct Manning	59
Confidence Intervals	60
Direct Manning Sensitivity	63
Manning Document	79
Summary	89
VI. Conclusions and Recommendations	90
Conclusions	90
Recommendations	96
Future Work	97
Bibliography	98
Appendix A: F-15 TFTW Maintenance Data Base	101
Appendix B: F-15 TFTW Operations Data File	149
Appendix C: Performance Summary Reports and Matrices	162
Appendix D: Failure Clocks	172
Appendix E: Spare Parts and Avionic Test Station Constraints	177
Vita	181
Vita	182

List of Figures

<u>Figure</u>	<u>Page</u>
1. LCOM and Moody Regression/Manpower Program Relationship . . .	5
2. LCOM Program and Data Relationship.	6
3. LCOM Simulation of Maintenance Activities	8
4. LCOM Maintenance Network.	11
5. Moody Regression Graph.	17
6. LCOM F-15 TFTW Maintenance Organization Structure	20
7. Sequence of Simulation Runs	31
8. Type I Hypothetical Steady State Behavior	35
9. Type II Hypothetical Steady State Behavior.	36
10. AFSC 531X3 (Structural Repair) Daily Manhours Used.	48
11. AFSC 531X3 (Structural Repair) Weekly Manhours Used	49
12. AFSC 423X3 (Fuel Systems) Weekly Manhours Used.	50
13. AFSC 431X1 (Flight Line Crew Chief) Daily Manhours Used . . .	53
14. Accomplished Aircraft Sortie Rate for Constrained Manpower and Unconstrained Parts/ATS Simulation	55
15. AFSC 431X1 (Flight Line Crew Chief) Autocorrelation Function.	57
16. AFSC 423X3 (Fuel Systems) Autocorrelation Function.	58
17. AFSC 431X1 (Flight Line Crew Chief) Direct Manning.	65
18. AFSC 431X1 (Phase Inspection) Direct Manning.	66
19. AFSC 531X3 (Structural Repair) Direct Manning	67
20. AFSC 426X2 (Jet Engine) Direct Manning.	68
21. AFSC 423X3 (Fuel Systems) Direct Manning.	69
22. AFSC 423X0 (Electrical Systems) Direct Manning.	70
23. AFSC 423X4 (Pneudraulics) Direct Manning.	71
24. AFSC 423X1 (Environmental Systems) Direct Manning	72

List of Figures (Continued)

<u>Figure</u>		<u>Page</u>
25.	AFSC 423X2 (Egress Systems) Direct Manning.	73
26.	AFSC 326X2B (Automatic Flight Control/Instruments) Direct Manning.	74
27.	AFSC 326X2A (Inertial Navigation System/Weapon Control) Direct Manning.	75
28.	AFSC 326X2C (Communications/Navigations/Electronic Counter Measures) Direct Manning.	76
29.	AFSC 326X1D (Automatic Test Station) Direct Manning	77
30.	AFSC 326X1C (Manual Test Station) Direct Manning.	78
31.	F-15 TFTW Total Manning Requirements.	81
32.	F-15 TFTW Manning Document.	82
A-1	LCOM Extended Form 11	102
C-1	Unconstrained Simulation Performance Summary Report	164
C-2	AFSC 326X2C On-Equipment Unconstrained Manpower Matrix.	166
C-3	AFSC 531X3 Off-Equipment Unconstrained Manpower Matrix.	167
C-4	Constrained Simulation Performance Summary Report	168
C-5	AFSC 326X2C On-Equipment Constrained Manpower Backorder Matrix.	170
C-6	AFSC 531X3 Off-Equipment Constrained Manpower Backorder Matrix.	171

List of Tables

<u>Table</u>		<u>Page</u>
I.	Function Code to AFSC Translation	23
II.	Daily Maintenance Requirements.	28
III.	AFSC 531X3 Manpower Constraint Calculation.	51
IV.	LCOM AFSC's That Exhibited Initial Transient Conditions . .	54
V.	AFSC 531X3 Direct Manning Computation	59
VI.	AFSC 531X3 Confidence Interval Computation.	61
VII.	End of Runway Crew Direct Manning Computation	62
VIII.	Accomplished Sortie Rate (ASR) and Flying Hours per Month (FHPM) versus Constraint Type and Scheduled Sortie Rate . .	63
IX.	AFSC Minimum Crew Manning	80
X.	Comparison of Simulation Strategies	92
A-I	LCOM AFSC Shredout Used in F-15 Data Base	105
D-I	Failure Clocks.	174
E-I	Spare Part Constraints.	178
E-II	Avionic Test Station Constraints.	180

Symbols and Abbreviations

AFB	Air Force Base
AFIT	Air Force Institute of Technology
AFM	Air Force Manual
AFSC	Air Force Specialty Code
AGE	Aerospace Ground Equipment
α	Type I Error
ASD	Aeronautical Systems Division
ASR	Accomplished Sortie Rate
ATS	Avionic Test Station
AUTO	Automatic
β	Type II Error
COM	Communications
E	Mutually Exclusive Probability
$E()$	Expected Value
ECM	Electronic Counter Measures
$E(U)$	Mean Value of Mann-Whitney
FC	Functional Code
FHPM	Flying Hours per Month
G	Nonmutually Exclusive Probability
H_a	Statistical Alternate Hypothesis
H_0	Statistical Null Hypothesis
INS	Inertial Navigation System
L	Interval or Lag Between Data Points
LCOM	Logistics Composite Model

Symbols and Abbreviations (continued)

LRU	Line Replaceable Unit
M	Direct Manning in Number of Men
MPIP	Maintenance Posture Improvement Program
M_s	Manpower Constraint in Number of Men
MSBMA	Mean Sorties Between Maintenance Action
N	Sample Size
NAV	Navigations
NRTS	Not Repairable This Station
P(R)	Probability That R Equals Some Integer Valued Number of Runs
PSR	Performance Summary Report
R	Number of Runs
ρ	Autocorrelation Coefficient
$\hat{\rho}$	Estimated Autocorrelation Coefficient
$\hat{\rho}(L)$	Estimated Autocorrelation Coefficient at Lag (L)
$\rho(L)$	Autocorrelation Coefficient at Lag (L)
RPV	Remotely Piloted Vehicle
σ^2	Population Variance
S_μ	Standard Deviation of the Mean
S_μ^2	Sample Variance of the Mean
TAC	Tactical Air Command
TFTW	Tactical Fighter Training Wing
U	Mann-Whitney Statistic
μ	Population Mean
U.E.	Unit Equipment

Symbols and Abbreviations (continued)

UHF	Ultra High Frequency
USAF	United States Air Force
V(U)	Variance of Mann-Whitney Statistic
W	Mann-Whitney Rank Value
WUC	Work Unit Code
\bar{X}	Sample Mean
X_t	Output Data Value at Time (t)
Z	Standard Normal Random Variable

I. INTRODUCTION

Estimation of the maintenance manpower necessary to support desired flying activities in various United States Air Force (USAF) organizations is a continuing problem. As new aircraft enter the inventory and as procedures change, there exists a recurring need for reliable estimates of the maintenance manpower necessary to support desired levels of flying activity efficiently. These estimates aid USAF managers in allocating maintenance manpower to new or existing flying units and insuring combat readiness.

Background

One method which has been used successfully to estimate these maintenance manpower requirements involves the use of the Logistics Composite Model (LCOM). The LCOM is a USAF computer simulation language designed to model USAF base level aircraft, maintenance, and support functions (Ref 33). Specifically, the model can be used to estimate maintenance manpower requirements for a USAF flying wing at specified levels of flying activity.

Two previous Air Force Institute of Technology (AFIT) theses have addressed LCOM estimation of maintenance manpower requirements: Green and Rumble constructed an LCOM simulation to evaluate the effects of alternative operational, maintenance, and supply policies on remotely piloted vehicle (RPV) maintenance manning (Ref 14:ii). Fritz and Yates used LCOM to simulate the interaction of the RPV, the launch aircraft, and the recovery helicopter (Ref 13:ii).

Tactical Air Command (TAC) used LCOM to estimate maintenance manpower requirements for their F-4, A-7, A-10, F-15, and F-16 aircraft

(Ref 23). However, TAC conducted the majority of these studies using a wartime operational environment and devoted little attention to a peacetime environment. Consequently, TAC suggested that AFIT students consider a peacetime LCOM study as a possible graduate thesis topic (Ref 29) and offered to make available an on-the-scene technical advisor to assist in making such a study meet TAC requirements.

Each of these previous LCOM studies simulated concurrent flying and maintenance activity. That is, aircraft maintenance was performed only on days of scheduled flying operations. In a wartime environment, this practice is acceptable since aircraft missions are scheduled seven days a week. However, this practice has one major drawback: if a high level of flying activity is scheduled, the aircraft maintenance organization may, at times, become overloaded with work. This causes a temporary decrease in flying activity until the maintenance organization clears out the backlogged work.

In a peacetime environment, flying operations are normally scheduled Monday through Friday. During high levels of flying activity, the maintenance organization continues to perform its functions on weekends in order to alleviate backlogged work. In this manner, the maintenance complex can usually keep stride with the weekly flying operations and the day to day level of flying activity remains fairly constant.

In this thesis, the authors use LCOM to model a peacetime flying environment. The model simulates a Monday through Friday flying schedule and a seven day maintenance work week.

Thesis Objectives

A need exists to expand F-15 LCOM estimation of maintenance manpower requirements to include the peacetime operational environment

(Ref 29). Furthermore, previous LCOM manpower studies have not emphasized statistical analysis of the output data.

In this thesis, the authors use LCOM to estimate the maintenance manpower requirements for an F-15 Tactical Fighter Training Wing (TFTW) with 72 unit equipment (U.E.) aircraft operating in a peacetime environment. In addition, they construct statistical confidence intervals around the resulting estimates of maintenance manpower requirements. Finally, the authors investigate the sensitivity of manpower requirements to variations in availability of aircraft parts and support equipment.

Thesis Scope

The LCOM peacetime environment is determined by TAC Training Syllabus Course Numbers F1500 B,I, and TX (Ref 30) requirements. These publications specify the flying training activity for F-15 pilot upgrade training which is the primary mission of a TFTW. The 58th TFTW located at Luke Air Force Base (AFB), Arizona currently uses these syllabi for all F-15 pilot upgrade training and is the base case for determining the peacetime F-15 LCOM operation and maintenance procedures.

Overview

The remainder of the thesis consists of five chapters. The LCOM chapter describes the Logistics Composite Model and Moody Manpower/Regression Programs. The Data Base chapter describes the maintenance and operations data base. The Methodology chapter lays the groundwork for this study's estimation of manpower requirements. The Analysis and Results chapter contains the manpower estimations and describes their sensitivity to variations in aircraft spare parts and support equipment.

Finally, the Conclusions and Recommendations chapter summarizes the thesis findings.

II. LCOM MANPOWER ESTIMATION

Rather than present a detailed description of the LCOM process, this chapter introduces simplified LCOM concepts which form a basis for the remainder of the thesis. Further detail concerning LCOM can be found in Drake (Ref 7 and 8) and Tetmeyer (Ref 33). The LCOM manpower estimation process uses the Logistics Composite Model, the Moody Regression Program, and the Moody Manpower Program (Ref 7, 8, and 33). The interaction of these computer programs produces a complete basic manning document for a USAF maintenance organization.

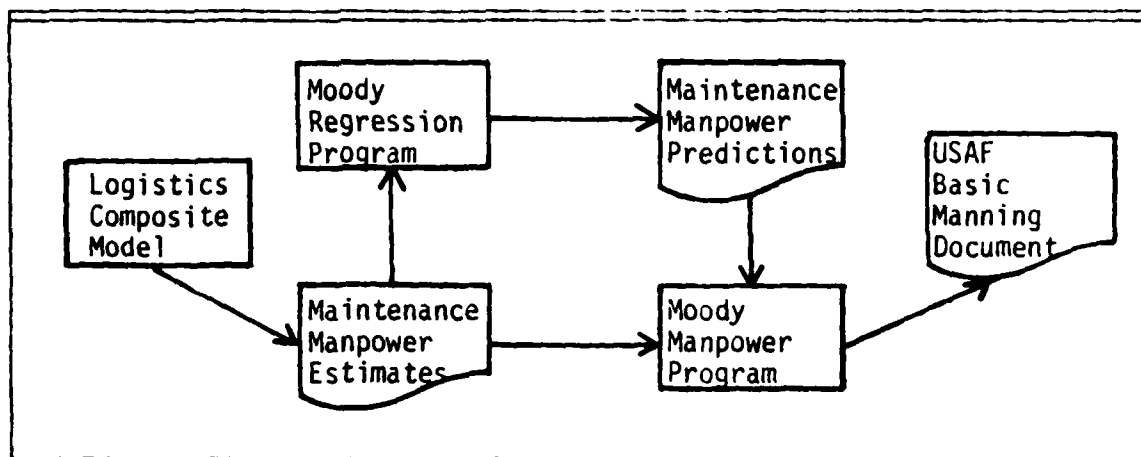


Figure 1. LCOM and Moody Regression/Manpower Program Relationship

Figure 1 depicts the interrelationship of LCOM and the Moody Regression/Manpower Programs. The Logistic Composite Model estimates maintenance manpower requirements for specified levels of flying activity. The Moody Regression Program uses these estimates and regression techniques to predict maintenance manpower requirements for a wide range of flying activity. The Moody Manpower Program uses the LCOM estimates and Regression predictions to produce a complete basic manning document for a USAF

maintenance organization. The following paragraphs describe each program in greater detail.

Logistics Composite Model

The LCOM uses three major computer programs to model aircraft flying operations, maintenance functions, and resource constraints. These programs are the preprocessor, main, and postprocessor programs (Ref 7: Chap. 1, p.2). The preprocessor program prepares aircraft operations and maintenance data for the main program. The main program simulates the interaction of aircraft operations, maintenance functions, and resource constraints and provides a statistical summary of the simulation results. The postprocessor program offers additional statistical output data.

Figure 2 illustrates the relationship between the three LCOM programs and the input/output data.

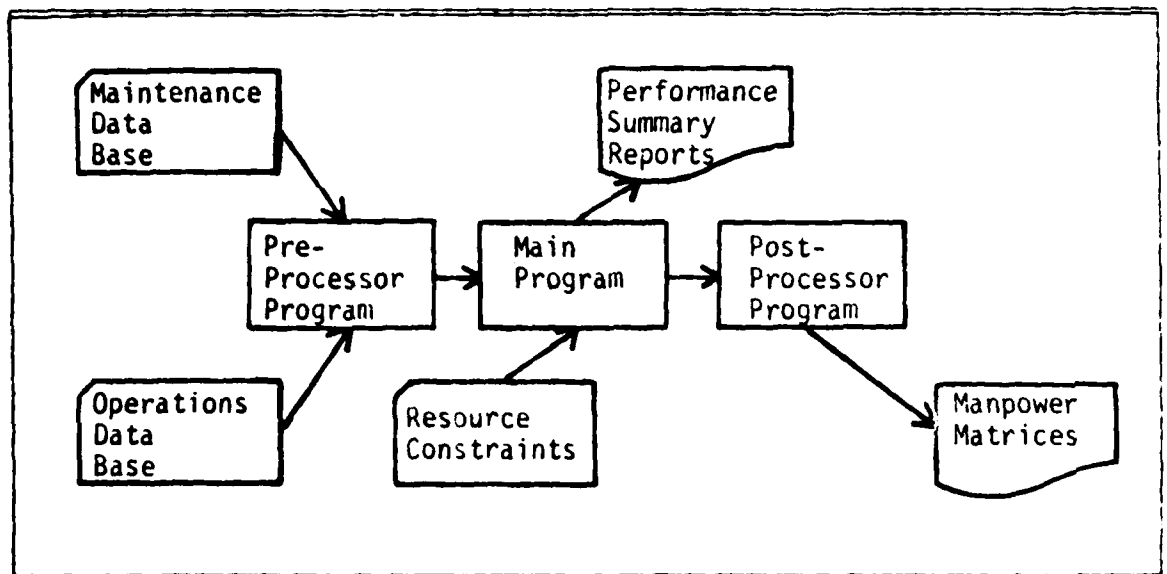


Figure 2. LCOM Program and Data Relationship

The operations and maintenance data bases represent the respective aircraft flying operations and maintenance characteristics. Resource

constraints consist of specified quantities of men by Air Force Specialty Code (AFSC), aircraft spare parts, and support equipment. The performance summary reports depict statistical summaries of the simulation results. Finally, the manpower matrices illustrate the daily distribution of manpower requirements (by AFSC).

Figure 3 shows how LCOM uses the flying operations schedule, maintenance functions, and resource constraints to simulate a sequence of maintenance activities. When the flying schedule calls for aircraft to start mission preparation, LCOM designates aircraft from the available aircraft pool for the mission. Each aircraft then processes through the preflight to postflight check blocks.

During this processing, LCOM uses men, spare parts, and support equipment as needed to perform maintenance functions. If all available manpower is already performing aircraft maintenance, LCOM delays the next mission until maintenance manpower is available. If the aircraft are ready for launch at their scheduled takeoff time, the missions fly for the specified mission length and then return for processing through the postflight check block. After postflight, LCOM places the aircraft in the available aircraft pool.

The LCOM also maintains a failure clock on each aircraft subsystem. These clocks use an exponential failure distribution to determine the number of sorties flown until corrective maintenance for their respective subsystems. Since LCOM does not simulate in-flight activity, it checks the failure clocks only during preflight and postflight. If the number of sorties flown equals a particular subsystem's clock value, LCOM lists that component as failed and ceases mission processing. The failed component then processes through the corrective maintenance block and uses

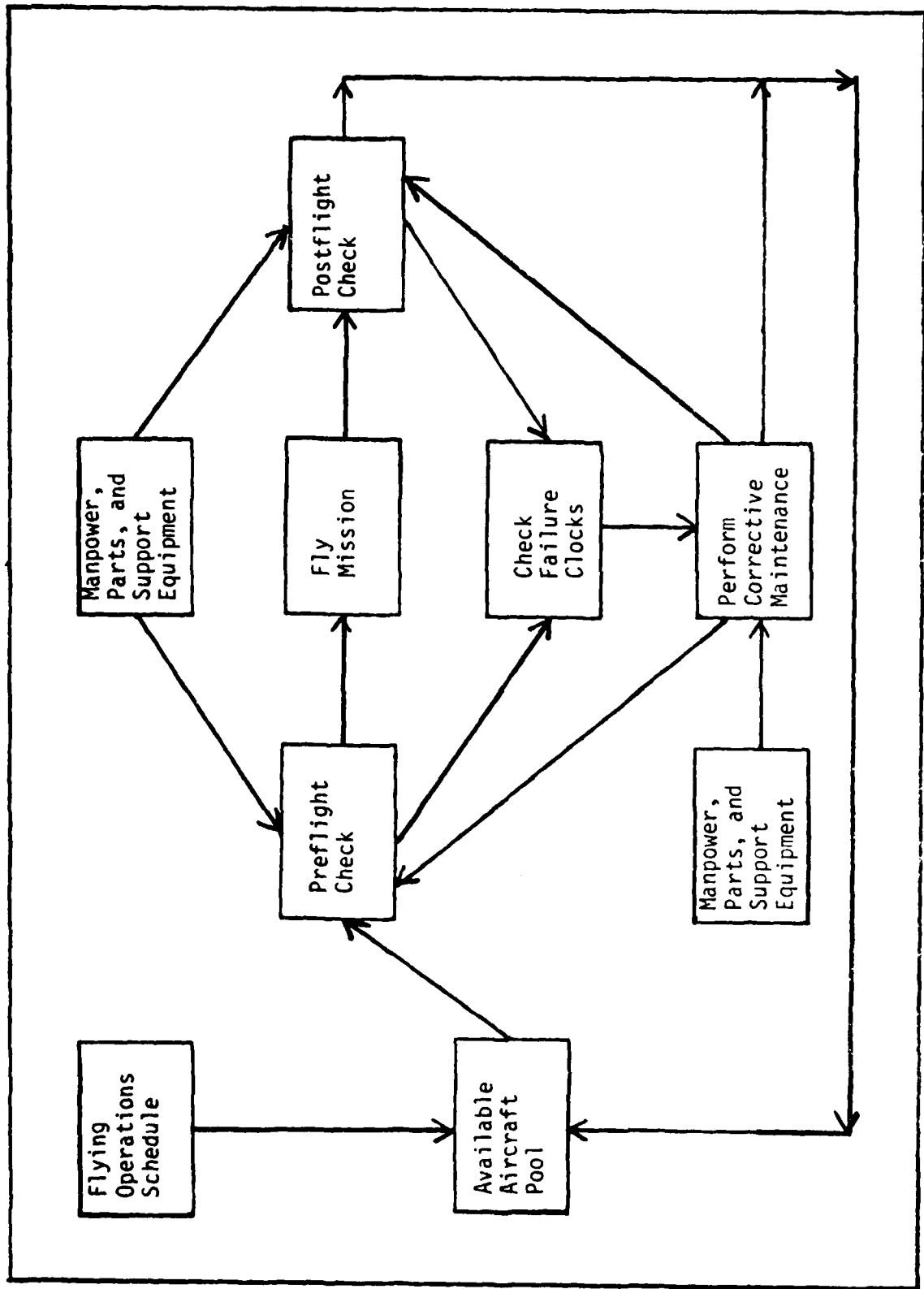


Figure 3. LCOM Simulation of Maintenance Activities

men, spare parts, and support equipment as necessary to perform the corrective maintenance. Upon completion of all corrective maintenance activity, LCOM allows the aircraft to continue with mission processing. However, if corrective maintenance delays an aircraft beyond its scheduled takeoff time, LCOM cancels the corresponding mission and returns the respective aircraft to the available aircraft pool.

The following description of LCOM's preprocessor, main, and post-processor programs more clearly defines the simulation process.

Preprocessor Program. The preprocessor program translates and organizes the maintenance and operations data bases for the main program. During data translation, the program scans the data base for inconsistencies and provides error diagnostic messages for data ambiguities (Ref 7: Chap. II, p.1). In some cases, the program makes computer logic assumptions when it finds minor data errors concerning user intentions and provides a message specifying the data ambiguity and corresponding program assumption. This feature prevents an unnecessary computer abort for minor data errors.

The LCOM maintenance data base consists of a weapon system's scheduled and unscheduled maintenance procedures, major components (parts), component failure frequencies, mean service and repair times, and resource (men, part, support equipment) requirements. This data represents the maintenance environment of an LCOM simulation (Ref 33:30).

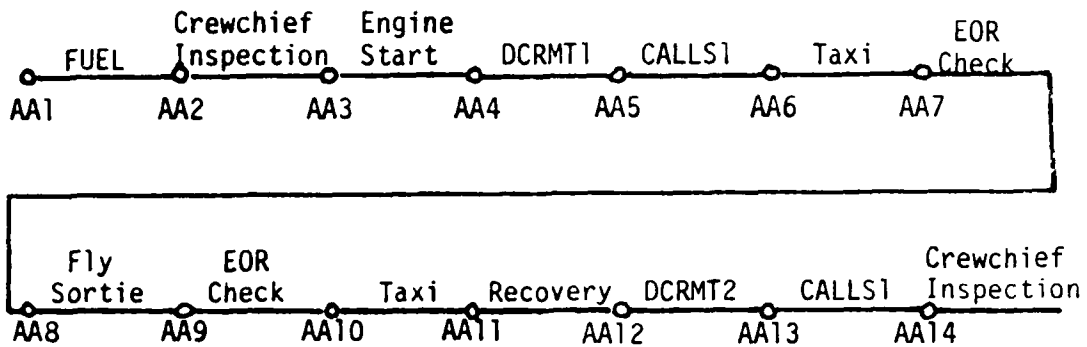
During data base formulation, the user graphically depicts the maintenance environment using LCOM networks (Ref 33:28). These networks define maintenance relationships within the data base. Figure 4 illustrates a simplified LCOM main network and corrective maintenance network for an aircraft mission.

In an LCOM network, tasks represent the scheduled and unscheduled maintenance procedures. These tasks are connected by network nodes (Ref 33:32). For example, nodes AA1/AA2, in Figure 4, define a FUEL task. The user also specifies the men (by Air Force Specialty Code), support equipment, and service or repair time necessary to complete a task. If a task resource is not available, LCOM will delay processing the aircraft mission beyond the guilty task until the constraining resource becomes available. For example, the FUEL task requires two men (AFSC 431X1), one fuel truck, and five minutes.

Every weapon system is composed of many major components or parts. The work unit code (WUC) manual for the particular weapon system numerically defines each of these parts. The LCOM maintains a failure clock for each major component in the maintenance network (Ref 33:36). For example, nodes M1/M2, in Figure 4, define the failure clock for the UHF radio (F63A00). The user assigns each failure clock a failure frequency parameter. He determines this parameter by analyzing failure rates of corresponding real life weapon system components. Mean sorties between maintenance action (MSBMA) is the most common parameter (Ref 33:56). In Figure 4, F63A00 has a MSBMA equal to 10 sorties. At the completion of an aircraft mission, the failure clock for each component in that mission main network advances one sortie.

The LCOM allows the user to specify the percent of time a major component fails before or after mission launch through the use of a clock decrement task (Ref 33:37). Nodes AA4/AA5, in Figure 4, define the clock decrement task prior to launch (DCRMT1) while nodes AA12/AA13 define the clock decrement task after launch (DCRMT2) for the UHF radio.

Main Mission Network:



UHF Radio Corrective Maintenance Network:

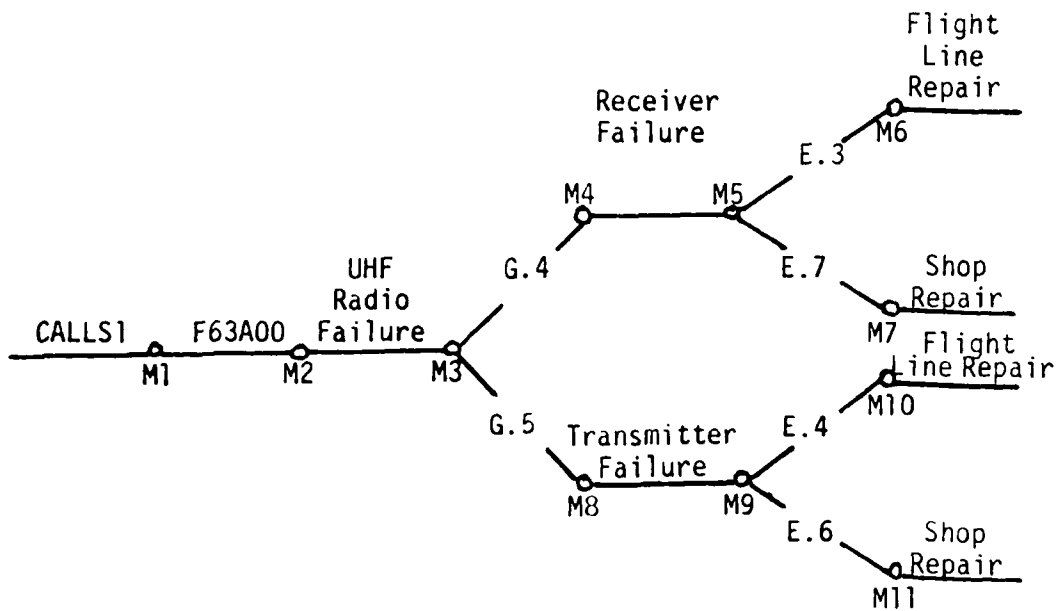


Figure 4. LCOM Maintenance Network

For example, DCRMT1 advances F63A00 three-fourth sortie and DCRMT2 advances F63A00 one-fourth sortie for a one sortie advance of the UHF radio clock by the completion of a processed aircraft mission.

Certain component failures in an LCOM network are not sortie related. In these cases, the failure clock uses a parameter other than MSBMA and the decrement task advances the failure clock by an adjusted amount independent of a sortie. One example is the gun on a tactical aircraft. Periodic gun maintenance is usually based on number of expended ammunition rounds. In this case, the failure clock uses rounds fired as a parameter and the decrement task advances the clock a certain number of rounds each mission sortie.

The LCOM interrogates a failure clock via a Call task (Ref 33:36). Nodes AA5/AA6 and AA13/AA14, in Figure 4, define the Call task (CALLS1) for the UHF radio failure clock (F63A00). Each time LCOM processes a mission through a Call task, the Model checks the corresponding clock to see if it has advanced to its MSBMA parameter. If the clock has advanced to this parameter, LCOM marks that component as failed and stops processing the aircraft mission. At the same time, the model begins processing the failed component through its respective corrective maintenance network. In Figure 4, LCOM processes the UHF radio through the UHF radio corrective maintenance network after F63A00 advances 10 sorties.

Within the corrective maintenance network, LCOM allows the user to specify which subsystem(s) caused the component failure and who fixes the subsystem(s). The model accomplishes this function with G and E probability distributions (Ref 33:62-65).

A G probability is a nonmutually exclusive probability that determines which subsystem(s) caused the component failure. Since the G

probability is nonmutually exclusive, either one or more than one subsystem can cause a component failure. In Figure 4, there exists a .4 probability that the receiver (nodes M4/M5) caused the radio failure, a .5 probability that the transmitter (nodes M8/M9) caused the radio failure, and an inferred .1 probability that both the receiver and transmitter caused the radio failure.

An E probability is a mutually exclusive probability that determines who fixes the broken subsystem(s). Since the E probability is mutually exclusive, Flight Line Repair (node M6) or Shop Repair (node M7), but not both, can fix the receiver. In Figure 4, there exists a .7 probability that Shop Repair fixes the receiver and a .3 probability that Flight Line Repair fixes the receiver.

If Shop Repair fixes the receiver, LCOM immediately generates a new part from supply stock and allows the aircraft mission to continue processing through its main network. After Shop Repair fixes the receiver, LCOM returns the repaired receiver to supply stock.

After constructing the LCOM networks, the user transcribes this information onto LCOM Extended Forms 11 (Ref 33:32-36). He then transfers this data to a computer card deck. This card deck is the completed maintenance data base. Figure A-1 in Appendix A contains a sample LCOM Extended Form 11.

The operations data base consists of the aircraft daily flying and maintenance schedules. These schedules form the operational scenario for an LCOM simulation.

The flying schedule contains the aircraft mission type, number of primary aircraft for each mission, mission takeoff time, mission cancel time, and flight duration. Missions with similar configurations, flight

time, preflight and postflight servicing, and maintenance crew requirements are grouped under a single LCOM mission type (Ref 33:16), since LCOM only simulates ground activity. As far as LCOM is concerned, flight time is that time which an aircraft is unavailable for maintenance.

The maintenance schedule specifies the number of spare aircraft per mission type, missions that fly more than once during the day (preflight to thru-flight), and number of phase inspections and aircraft washes.

The user designs the operational scenario for a specified level of flying activity. This level of activity is usually designated as aircraft sortie rate. Equation (1) defines sortie rate.

$$\text{sortie rate} = \frac{\text{sorties/day}}{\text{aircraft U.E. size}} \quad (1)$$

The U.E. size is the number of authorized aircraft assigned to a unit. Sorties per day may be either scheduled sorties or accomplished sorties. During the planning stage, the user bases the operational scenario on a scheduled aircraft sortie rate. After an LCOM simulation, the user computes an accomplished aircraft sortie rate.

The user records the completed operational scenario onto LCOM Forms 20 (Ref 33:17) prior to transferring the information to a computer card deck. The punched card deck represents the operational data base. Appendix B contains three operational scenarios based on scheduled sortie rates of .43, .74, and 1.0.

Main Program. The main program simulates maintenance and operations data base interaction. During this interaction, the main program uses available resources to prepare scheduled maintenance and flying

activity. The program processes each mission type through its respective main network while simulating all required maintenance tasks.

The preprocessor translated maintenance data base contains unconstrained maintenance manpower, aircraft spare parts, and support equipment. When these resources are left unconstrained, the main program has little difficulty complying with the operations data base's scheduled sortie rate provided sufficient time is allowed between scheduled missions for postflight and preflight maintenance tasks. If resources are constrained, the main program delays mission types, as necessary, when men, spare parts, and/or support equipment are unavailable. If a mission delay exceeds its cancel time, the program cancels the guilty mission. To constrain these resources, the user submits a series of computer cards to the main program prior to simulation. Each card specifies the number of maintenance men by AFSC and work shift, quantity of spare parts by WUC, and/or quantity of support equipment available during the LCOM simulation (Ref 33:123).

After simulation, the main program provides statistical data in the form of Performance Summary Report (PSR). These reports include the number of flying hours, number of sorties requested, number of sorties accomplished, manhours (by AFSC) used, manhour utilization rate (by AFSC), and parts (by WUC) consumed, generated, or backordered (Ref 7, Chap. II, p. 3). The user specifies the interval and number of PSR's desired prior to simulation. Figures C-1 and C-4 in Appendix C contain sample PSR's. The PSR's are used to evaluate the effects of main program simulation. For example, the user can compare scheduled and accomplished sortie rates to determine the effect of resource constraints on the simulation.

Postprocessor Program. The postprocessor program produces manpower matrices for each AFSC. These manpower matrices depict how an AFSC workload varies with the time of day (Ref 33:121). Figures C-2, C-3, C-5, and C-6 in Appendix C contain examples of on-equipment, off-equipment, and backorder matrix printouts for various AFSC's.

These on/off equipment printouts display the number of times during a simulated period that the number of people shown in the leftmost vertical axis are working at the time of day shown on the horizontal axis. This time is in half hour increments.

For instance, the on-equipment matrix for AFSC 326C2, in Appendix C, shows 11 cases in which two people are needed between 0330 and 0400.

The backorder matrices indicate which work shifts require additional personnel in order to increase sorties accomplished.

Moody Regression Program

The Moody Regression Program allows the user to estimate maintenance manning requirements for a wide range of flying activity (Ref 33:126). The user provides the program with three sets of LCOM estimated manning requirements (by AFSC) for an operational scenario based on three different scheduled sortie rates. The Moody Regression Program uses this data to compute first and second order regression equations for maintenance manning (by AFSC) as a function of flying hours per month. The program compares these ordered equations for goodness of fit and chooses that equation, for each AFSC, with the best fit. These equations can be used to graphically depict the information as shown in Figure 5.

Moody Manpower Program

The Moody Manpower Program produces a complete basic authorization document for a maintenance organization (Ref 33:129). The program

requires a unit's aircraft U.E. size and flying hours per month from the LCOM simulation, regression equations from the Moody Regression Program, the minimum crew size for each AFSC, and the organization's Major Command supervision and overhead requirements. The resultant document defines the organization structure by Air Force Functional Code (FC) and provides manning requirements by AFSC grade level, for each FC.

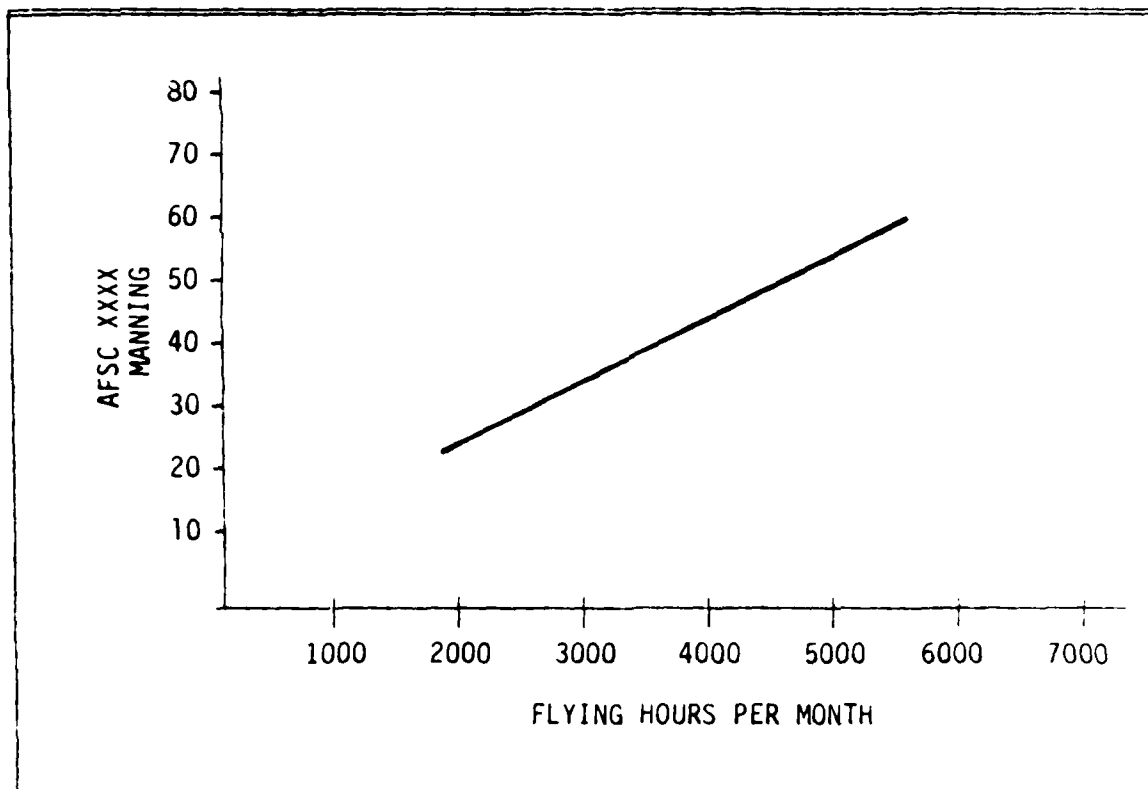


Figure 5. Moody Regression Graph

Summary

This chapter established the basic concepts used in the LCOM manpower estimation process. These concepts included the three LCOM programs, the Moody Regression Program, and the Moody Manpower Program. The next

chapter uses these concepts to describe the F-15 peacetime maintenance data base, operations data base, and resource constraints used in this study.

III. DATA BASE

This chapter contains a discussion of the maintenance and operations data bases and resource constraints used in the F-15 TFTW peacetime Logistics Composite Model. The description includes pertinent assumptions and the maintenance organization structure.

Maintenance Data Base

The maintenance data base is a modified version of maintenance data used in the ASD F-15/F-16 Wartime Study (Ref 5). Revisions include maintenance of TF-15 aircraft and simplified flight line maintenance networks with reduced armament and electronic counter measure requirements. Pages 107 through 149 in Appendix A contain a computer listing of the revised data base. The following paragraphs discuss these revisions in terms of the maintenance organization structure, scheduled and unscheduled maintenance procedures, repair/service times, weapon system components, and failure parameters.

Maintenance Organization Structure. Figure 6 illustrates the maintenance organization structure that supports the LCOM F-15 TFTW. This structure is similar to the 58th TFTW maintenance operation. The LCOM estimates those functional codes (FC's) designated by an "L" in Figure 6; the Moody Manpower Program produces a basic authorization document for all FC's depicted in the organization structure. Table I translates these FC's into AFSC's for LCOM estimated manpower and gives their work descriptions.

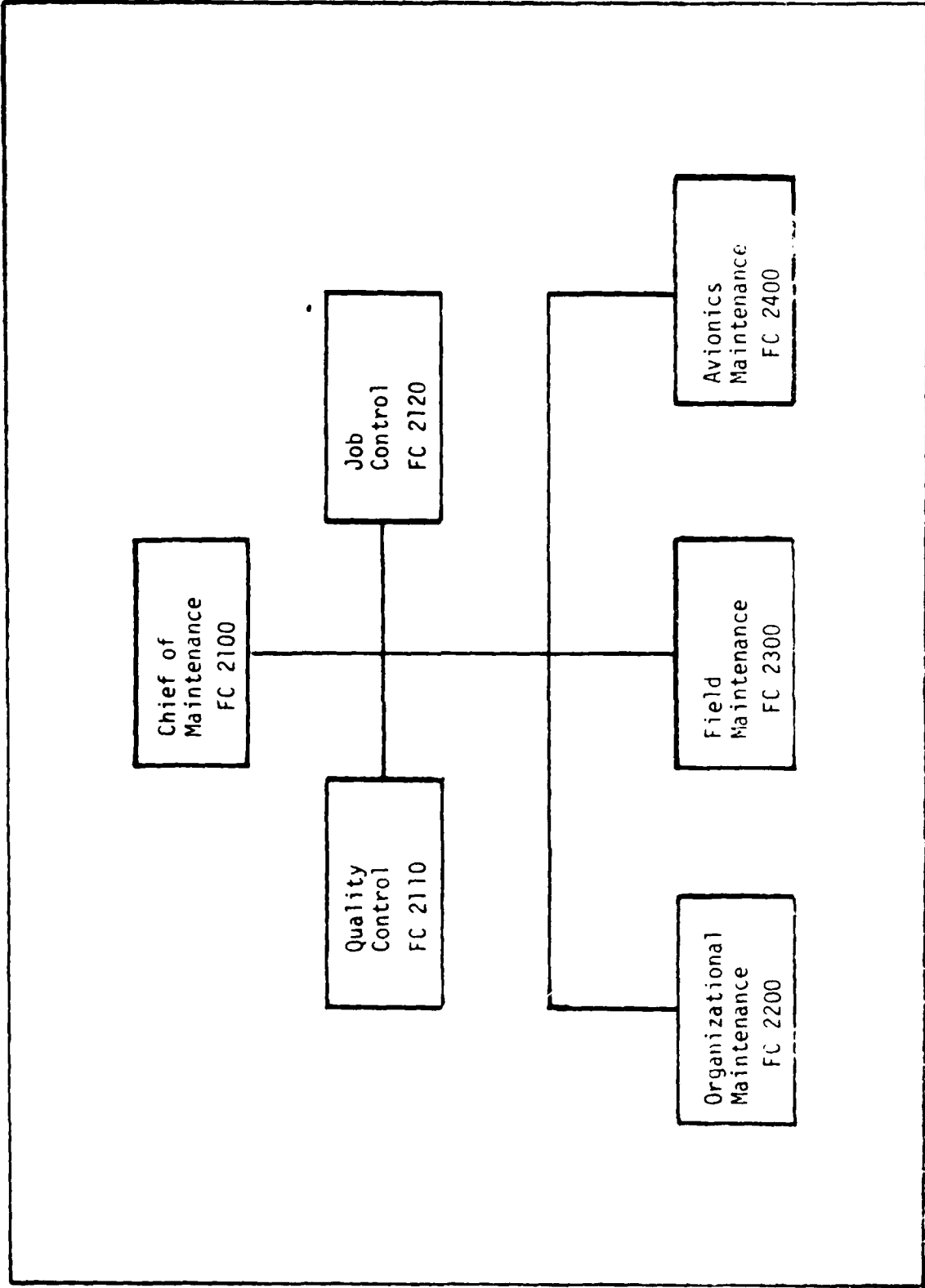


Figure 6. LCOM F-15 TFW Maintenance Organization Structure

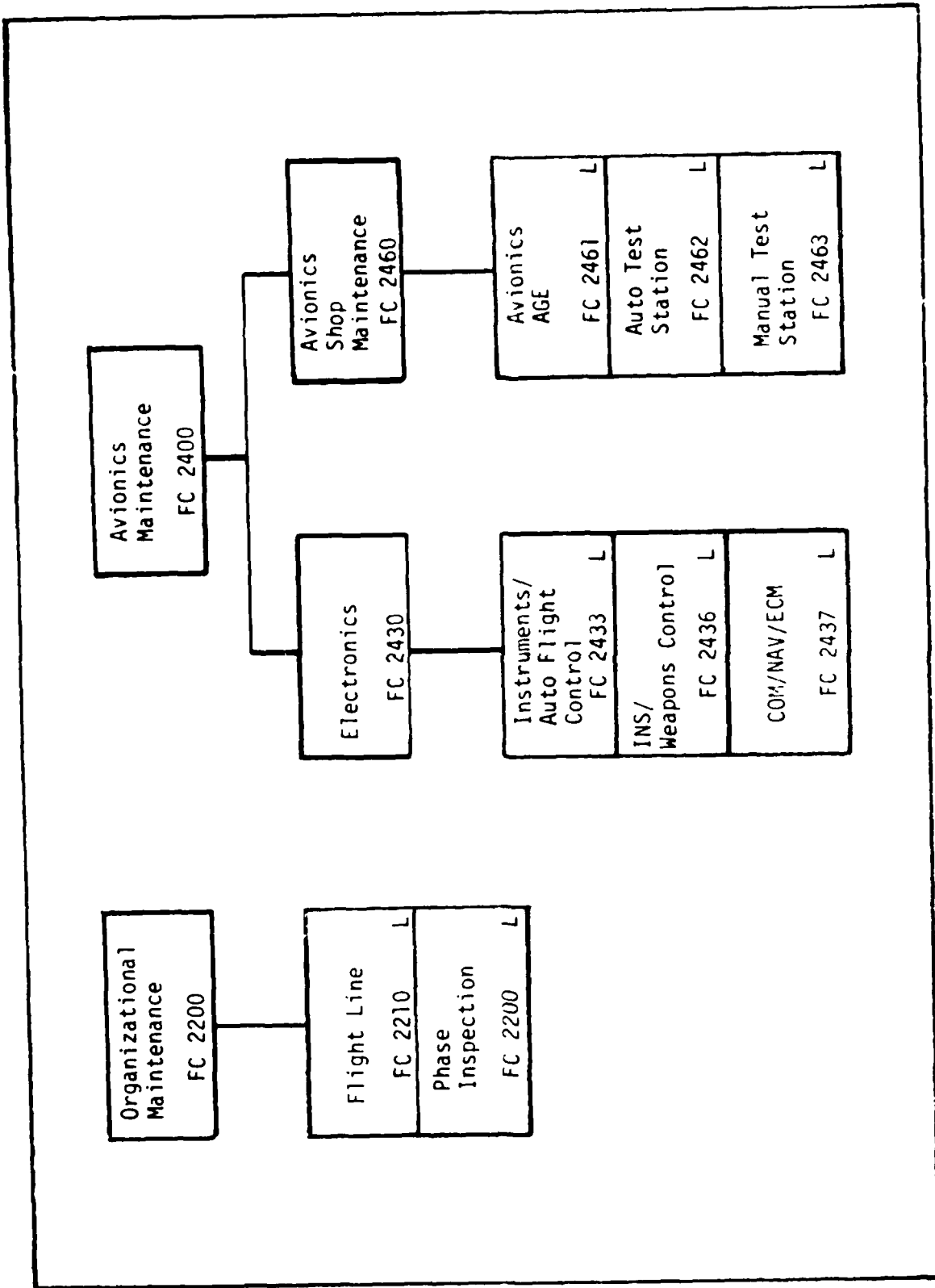


Figure 6. LCOM F-15 TFW Maintenance Organization Structure (Continued)

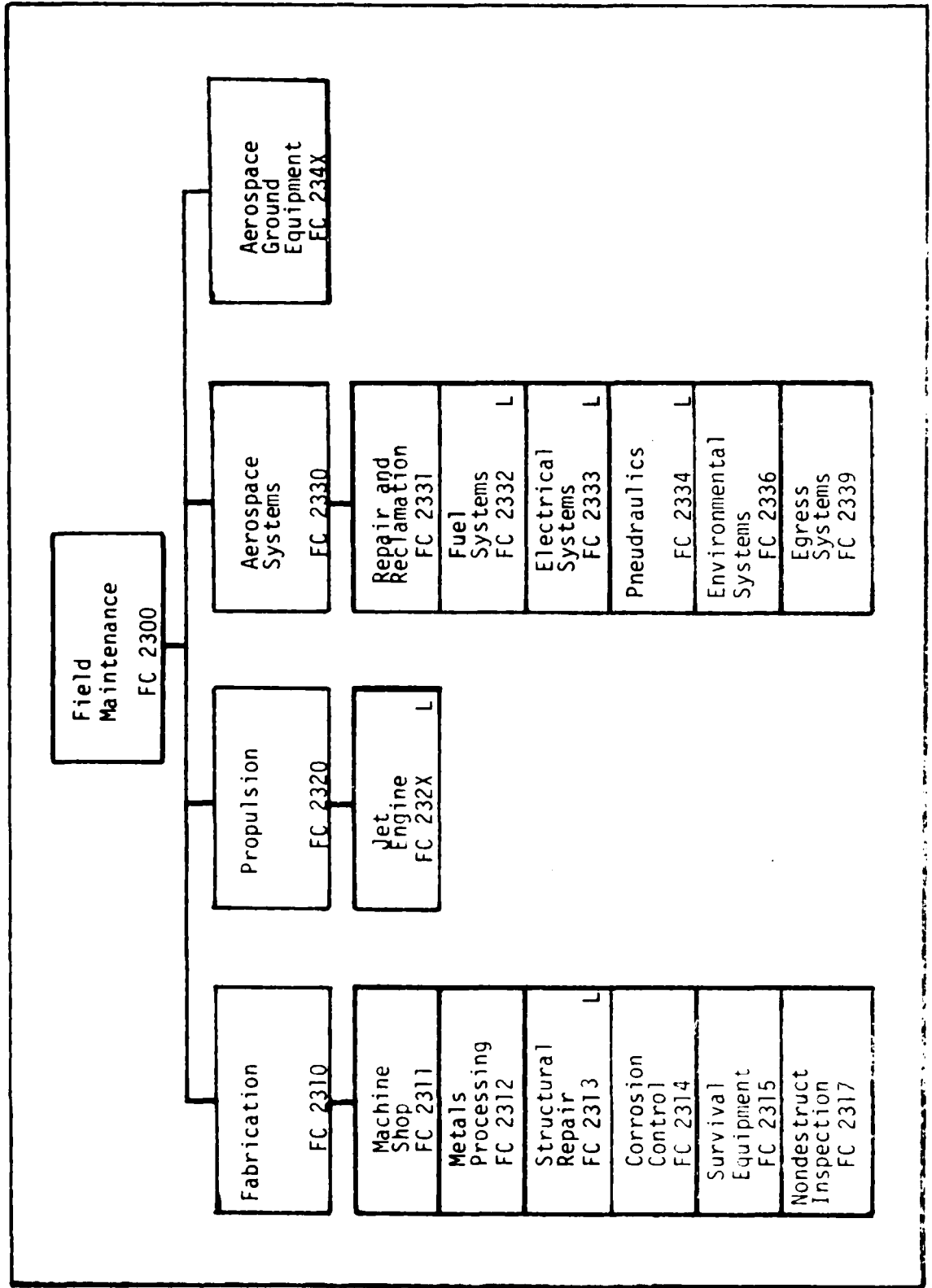


Figure 6. LCOM F-15 TFTW Maintenance Organization Structure (Continued)

Table I
Function Code to AFSC Translation

FC	WORK DESCRIPTION	AFSC
22XX	Flight Line Crew Chief/ Phase Inspection	431X1
2313	Structural Repair	531X3
232X	Jet Engine Shop/Jet Engine Flight Line/Test Cell	426X2
2332	Fuel Systems	423X3
2333	Electrical Systems	423X0
2334	Pneudraulics	423X4
2336	Environmental Systems	423X1
2339	Egress Systems	423X2
2433	Automatic Flight Control/Instruments	326X2B
2436	Inertial Navigation System/Weapon Control	326X2A
2437	Communications/Navigations/Electronic Counter Measures	326X2C
2462	Automatic Test Station	326X1D
2463	Manual Test Station	326X1C

Scheduled and Unscheduled Maintenance. The wartime data base contained maintenance procedures peculiar to a wartime environment. With the technical assistance of the TAC advisor, the authors deleted wartime procedures and added, when necessary, the appropriate peacetime maintenance functions. The resulting LCOM networks represent those scheduled and unscheduled maintenance procedures performed by the 58th TFTW located at Luke AFB, Arizona. These modifications are based on an operational audit of maintenance procedures at Luke AFB (Ref 22) and TAC coordinated peacetime maintenance assumptions (Ref 16, 17, and 27). These assumptions consist of the following:

- The LCOM simulation models maintenance procedures of a 72 U.E. TFTW operating under TAC Training Syllabus course numbers F1500 B, I, and TX (Ref 30).
- Both F-15A and TF-15A aircraft maintenance are modeled.
- The 20mm gun is loaded with 1,000 rounds of ammunition and reloaded after two DART missions are flown. This is the only "live" ammunition used in the simulation.
- Ground aborted missions are not rescheduled since course flying requirements have a nine percent refly factor in establishing the Flying Training Program.
- The wing maintenance activity is organized per AFM 66-1 (Ref 4) and does not reflect Maintenance Posture Improvement Program (MPIP) reorganization.
- The F-15/TF-15 phase inspection procedures performed by the 58th TFTW at Luke AFB are used in scheduled maintenance networks.

- Not reparable this station (NRTS) components use a 15 day resupply time.
- On/off equipment maintenance is available 24 hours a day and seven days a week.
- On-equipment failure of a component is corrected by a remove and replace task when parts are available.
- Cannibalization is initiated when spare parts are constrained and a removed component cannot be repaired in time to accomplish a mission.
- End-of-runway teams are always available during scheduled flying periods.

Maintenance Repair/Service Time. Maintenance task repair and service times are based on interviews with the respective work center supervisors at Luke AFB, Arizona (Ref 1). The data base listing in Appendix A depicts these times and the respective maintenance tasks.

Weapon System Components. The authors used the F/TF-15A Series Work Unit Code (WUC) Manual to describe all weapon system components (parts) (Ref 36). A WUC is a five character code (11XXX-99XXX) that identifies each weapon system part. The maintenance data base describes each F-15 part at the two-four significant character WUC level. This description is equivalent to the line replaceable unit (LRU) level.

Component Failure Parameters. The maintenance data tape (Ref 1) for the 58th TFTW at Luke AFB, defines failure parameters for components in the corrective maintenance networks. This tape contains maintenance

data for the September 1975 - February 1976 time period and includes 2132 F/TF-15 flying sorties and 2867 flying hours.

Tetmeyer, Moody, and Nichols (Ref 32 and 33) describe the detailed procedures necessary to extract useful maintenance data from this tape. They also illustrate the method to compute the LCOM failure parameters (MSBMA and G/E Probabilities).

The data base listing in Appendix A illustrates the G and E probabilities for each corrective maintenance network. Appendix D contains each component failure clock, its corresponding MSBMA, and decrement value. The decrement values come from the F-15/F-16 Wartime Study (Ref 5).

Operations Data Base

The F-15 peacetime operations data base contains the flying training schedule and maintenance wash and phase inspection schedules and supports a 72 U.E. TFTW consisting of 48 single seat (F-15) and 24 tandem seat (TF-15) aircraft. The following TAC coordinated assumptions (Ref 16, 17, and 27) were used during construction of the operational scenarios contained in Appendix B.

- TAC Training Syllabus Course Numbers F1500 B, I, and TX define the unit's flying requirements (Ref 30).
- The modeled TFTW conducts only a conversion and air-to-air training program.
- Flying schedule sortie length and variance are derived from the training syllabi.

- Flying training is not scheduled on weekends or holidays.
- Night flying is scheduled as training requires.
- Aircraft scheduled for morning flights are thru-flighted as required to meet mission requirements.
- Aircraft missions delayed longer than two hours are cancelled.
- Only scheduled and spare mission aircraft are preflighted each day.
- Preflight inspections for each aircraft are completed at least two hours prior to scheduled takeoff.

Flying Schedule. This study uses three flying schedules to obtain scheduled sortie rates of .43, .74, and 1.0. Each schedule contains a 13 percent over schedule for maintenance cancellations and a nine percent over schedule for rescheduled training requirements. The .74 sortie rate establishes the base line flying schedule while the .43 and 1.0 sortie rates allow for decreased or increased demands in F-15 pilots. The schedules use a 65/35 ratio of F-15 versus TF-15 aircraft.

The flying schedules involved two basic mission types: Conversion and Air-Air missions. Conversion missions familiarize the pilot with basic F-15 aircraft performance characteristics. Air-Air missions contain tactical air intercept, combat, and gunnery maneuvers. All three flying schedules are based on a 30/70 ratio of Conversion versus Air-Air missions.

Maintenance Schedule. The maintenance schedule specifies the number of spare aircraft, thru-flights, washes, and phase inspections. In this study, the following conditions apply: 52 percent of scheduled aircraft are thru-flighted each day; airframes receive a wash every 75 hours and a phase inspection every 50 hours; spare aircraft are based on 10 percent of scheduled sorties.

Table II summarizes the daily maintenance requirements for the three operational scenarios used in this study.

Table II
Daily Maintenance Requirements

	Scheduled Sortie Rate		
	.43	.74	1.0
Scheduled Aircraft	21	35	47
Scheduled Sorties	31	53	72
Spare Aircraft	4	6	8
Hours per Airframe per Month	15	25	33

Resource Constraints

During the LCOM simulation, the authors constrain selected spare parts and support equipment to measure the effects on accomplished sortie rate and maintenance manpower. These constraints reflect the quantity of spare parts and support equipment available from base supply stock.

This study uses part supply levels found in the F-15/F-16 Wartime manpower Study (Ref 5:60-64). These quantities reflect the most current information on part levels. Appendix E lists the constrained spare parts by WUC and the corresponding supply quantity.

The only support equipment constrained during simulation are avionic test stations (ATS's). Maintenance personnel use these stations to troubleshoot aircraft avionic components. There are three ATS types in the maintenance data base. Appendix E lists each type, the job description, and constrained quantity.

Summary

The F-15 TFTW maintenance and operations data bases and resource constraints were discussed in this chapter. The discussion included pertinent TAC coordinated assumptions and the modeled maintenance organization structure. The next section contains the procedures used in this study to simulate peacetime maintenance and operations data base interaction and analyze the resultant manpower estimates.

IV METHODOLOGY

In this chapter, the authors discuss the concepts and procedures employed in the development of manning estimates using the LCOM model, and the analyses of results obtained. Specifically, this includes an explanation of the sequence of simulation runs performed, the use of the resulting data to compute manpower estimates, and the procedures employed to validate the model. Further, the authors discuss the statistical concepts used to define steady state for the model and to construct confidence intervals for the manpower estimates. Finally, the use of the Moody Regression and Moody Manpower Programs to investigate the sensitivity of manning requirements to parts/ATS constraints is explained.

Sequence of Simulation Runs

In order to estimate manpower requirements for a particular combination of sortie rate and parts/ATS constraints, a series of LCOM runs was performed as shown in Figure 7. To investigate the sensitivity of manpower requirements to various constraints and sortie rates, this sequence was repeated for each of the nine possible combinations of the three sortie rates (.43, .74, and 1.0) and the three levels of resource constraints (unconstrained parts/ATS, constrained parts and unconstrained ATS, and constrained parts/ATS).

Determination of Manpower Requirements

The LCOM determined manpower requirements are based on the total manhours used for each AFSC. After running the model with unconstrained manpower as indicated in Figure 7, it is necessary to constrain the number of men available in the model in order to estimate the actual number

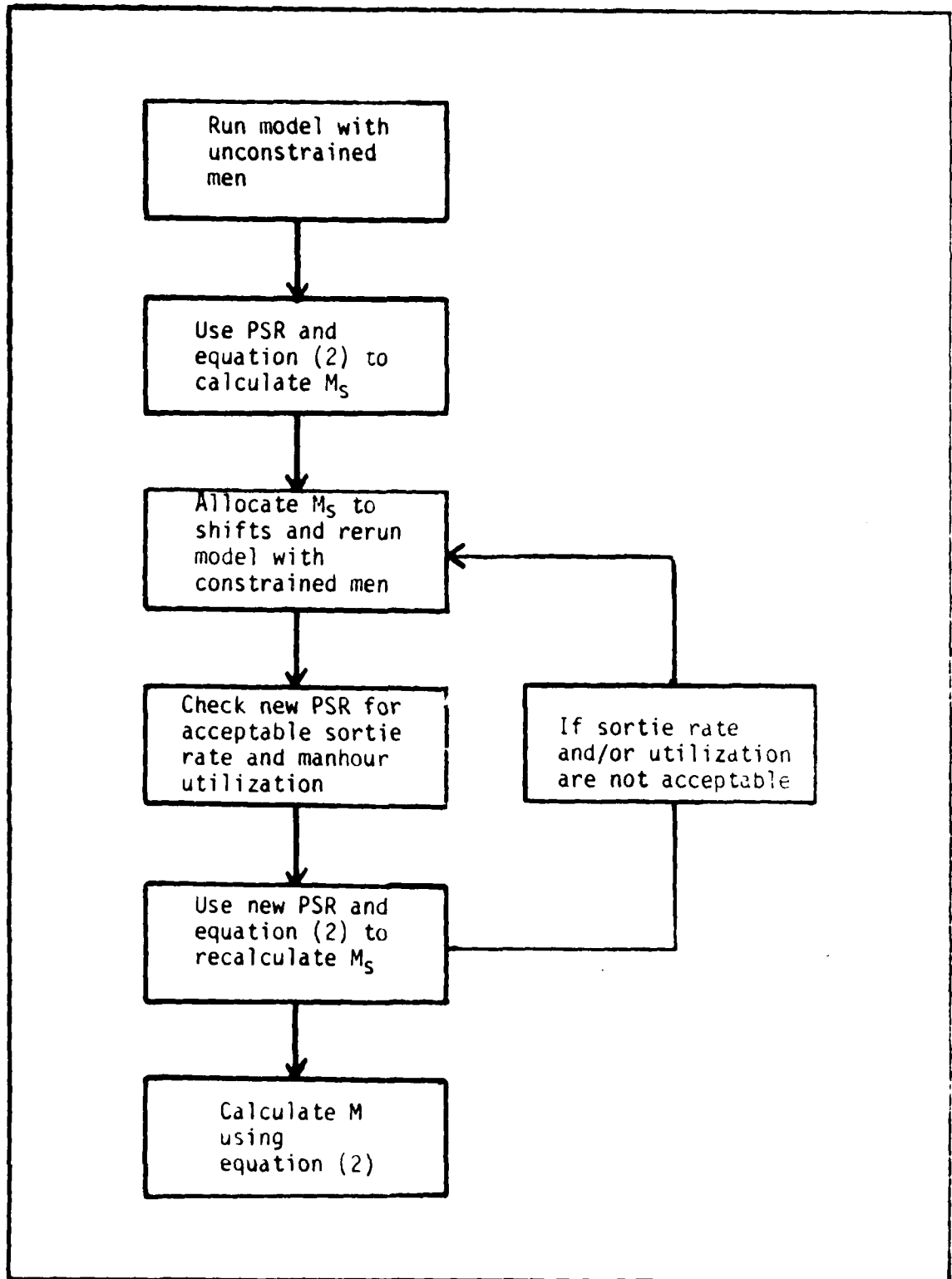


Figure 7. Sequence of Simulation Runs

of men required to support a specified sortie rate. Equation (2) is used to convert each AFSC's total manhours used into these manning constraints.

$$M_s = \frac{\text{Total Manhours Used}}{\left(\frac{\text{Utilization}}{\text{Factor}}\right) \left(\frac{\text{Number}}{\text{of Days}}\right) \left(\frac{\text{Shift}}{\text{Length}}\right)} \quad (2)$$

In this equation, the total manhours used for each AFSC are obtained from the simulation's performance summary report (PSR). The utilization factor is the fraction of manhours available for direct work; in this case, the Air Force standard is 0.6 (Ref 2: Chap. 6, p. 29). Shift length is eight hours. The M_s then becomes the average number of men available each day for each AFSC (Ref 33:121). This number is allocated over the three shifts using the manpower matrices as a guide. The resulting shift manning is incorporated into the model as a constraint on manpower. Certain AFSC's require a minimum crew size to perform their maintenance functions. In cases where M_s is less than the daily minimum crew manning, the minimum crew size is assigned to each shift as the constraint on manpower. The simulation is then rerun to determine what change has occurred in the accomplished sortie rate, as defined by equation (1), and to ensure that the resulting manhour utilization rate has not exceeded 0.6. If the accomplished sortie rate and/or manhour utilization rate are unacceptable, the process is repeated as indicated in Figure 7.

The M_s resulting from this procedure, however, is not the actual number of men required because it does not consider the fact that each individual normally works five days a week, takes leave, and may otherwise be legitimately absent from work. The Air Force standard for actual hours worked per man per month is 144 hours (Ref 3:7-8). Equation (3) is used to convert M_s into Direct Manning (M). Direct Manning

represents the final manpower estimate for each AFSC in the LCOM simulation.

$$M = \frac{M_S(\text{work days/month})(\text{shift length})}{144 \text{ hours/man/month}} \quad (3)$$

Model Validation

If a simulation model is to be of use in making predictions, then those predictions must be shown to be correct or at least as good as predictions made using some other method. Emshoff and Sisson state that the best possible evidence of such validity is that the model has made satisfactory predictions in the past (Ref 10:204). In this respect, the general LCOM model has been proven valid by numerous previous studies of other weapons systems and scenarios. There was, however, no direct counterpart to the present study, and for this reason a further investigation of validity was conducted.

A first-time model, such as the one under consideration here, cannot be validated completely until the system it represents actually comes into existence. Until that time, the model's validity can only be evaluated in terms of credibility. Emshoff and Sisson suggest that such validation consists of debugging the program, checking that key subsystem models predict their part of the world well (using historical data), and that knowledgeable individuals agree to the reasonableness of the model structure and output or face validity (Ref 10:206). The authors used these criteria in conjunction with a more detailed debugging procedure, recommended by Tetmeyer, to develop the following validation procedure (Ref 33:111, 113).

A 28 day unconstrained simulation was run, with complete diagnostics, and the output checked to insure that: all phases and scheduled

inspections were accomplished as well as a high percentage of scheduled sorties, there were no unsatisfied demands for personnel, the number of repairable generations for each item in the shop repair summary was equal to the number of units demanded for the same item in the supply summary, and, finally, there were no cannibalizations, supply backorders or unsatisfied demands (anything else would have indicated a data error) (Ref 33:113).

Next, a 28 day simulation was run with each of the three flying schedules. In each case, all unscheduled maintenance was removed from the model, thus preventing any equipment or aircraft breakdowns. In each case, all scheduled sorties were in fact accomplished, demonstrating the feasibility of the flying scenarios.

Finally, the MSBMA assigned from historical data to each failure clock was compared to the MSBMA actually observed during a 28 day run of the simulation. In most cases the variation was low; those few cases that did show a large variation in MSBMA had a large assigned MSBMA (the failure should seldom occur). In these cases small numerical differences between observed and predicted numbers of failures resulted in large changes in MSBMA. In light of the negative exponential distribution of failures in the model, and the short time span of the test simulation, variations of this sort were not unexpected. Further, as noted by Emshoff and Sisson, variations in the distribution of infrequent events are not normally significant when the model is intended to predict the behavior of some larger system (Ref 10:205). Consequently, the authors saw no reason to doubt the validity of the model based on its stochastic behavior.

The model was also tested for face validity. Emshoff and Sisson state that face validity exists when knowledgeable individuals agree to the reasonableness of the model structure and output (Ref 10:204). In this case, Lowell and Moody examined the model and output and found it, in their opinion, to be reasonable (Ref 24).

Steady State

In any simulation model, one of the important questions the researcher must answer is, "How long must the simulation be run until the parameters of interest (in this study sortie rate and manhours used) reach typical values for the system, or steady state?" Steady state, of course, is a relative condition depending on the specific parameters being considered and the degree of accuracy desired from the simulation. For instance, the time series output of a hypothetical simulation might appear like that shown in Figure 8.

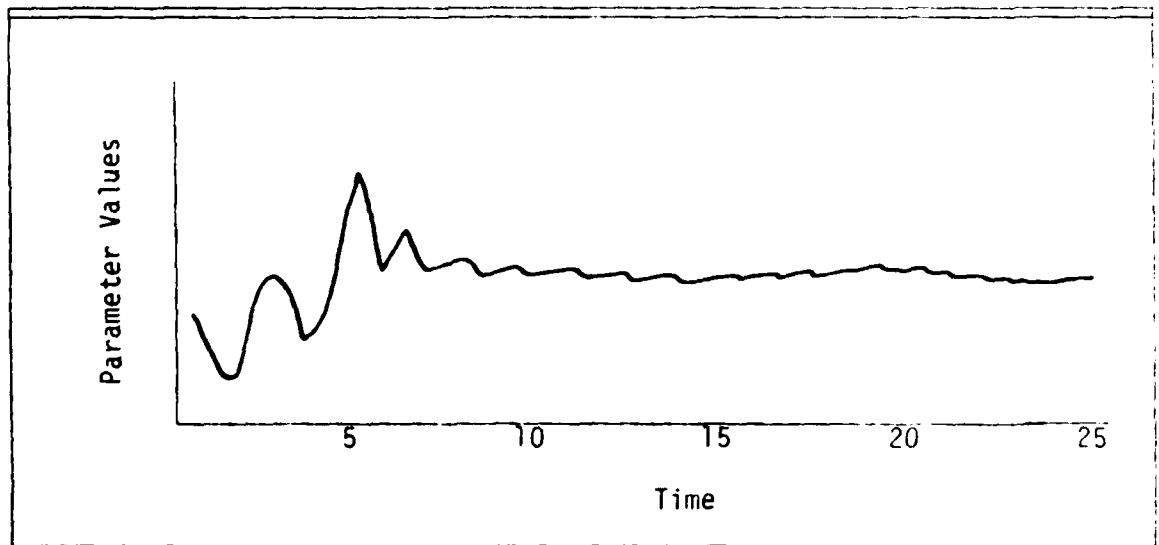


Figure 8. Type I Hypothetical Steady State Behavior

In this particular hypothetical example, after time period 10 the value of the output parameter is reasonably constant. In such a situation, the user could confidently accept data after that time as being typical of the system being modeled.

Unfortunately, many simulations do not exhibit the type of behavior just described. The values of output parameters may increase or decrease over time, indicating that they do not have a typical value within the system being simulated. Another possibility is that the value of the output may fluctuate considerably from one observation to the next; yet, in viewing such data over time the individual data points show no particular trends; but instead, appear to be randomly distributed about some central value. Such a case is illustrated in Figure 9. A closer

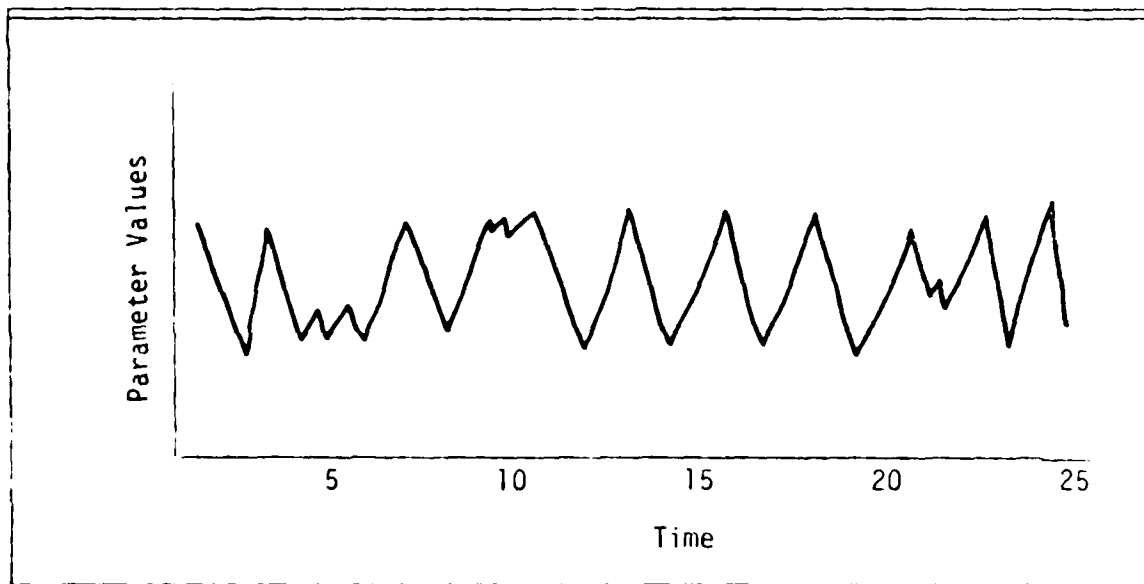


Figure 9. Type II Hypothetical Steady State Behavior

examination of the two situations depicted in Figures 8 and 9 reveals that the graph in Figure 8 after period 10, is actually similar to the situation depicted in Figure 9 except for the amplitude of the fluctuations.

Some Criteria for Steady State

Tetmeyer states in his discussion of the LCOM model that steady state is reached when successive thirty day manhour totals for each AFSC do not vary more than five percent (Ref 33:112). Of course, this is just another way of stating how much error the user is willing to accept.

As noted by Conway:

"...it is also important to recognize that equilibrium is a limiting condition which may be approached but never actually attained. This means that there is no single point in the execution of a simulation experiment beyond which the system is in equilibrium. The difference between the temporal and limiting distribution presumably decreases with time and one seeks a point beyond which he is willing to neglect the error that is made by considering the system to be in equilibrium" (Ref 6).

Emshoff and Sisson mention a procedure similar to that used by Tetmeyer, but go on to discuss other methods which the authors of this study felt were possibly more appropriate (Ref 10:192). Specifically, they state that steady state has been reached when successive output values vary randomly about some typical value; or similarly, when successive observations of the time series output are statistically indistinguishable (Ref 10:190).

The procedure described by Tetmeyer for identifying steady state is in common usage among LCOM practitioners. In an effort to reduce computer running time, the authors elected to investigate the feasibility of determining steady state by comparing probability distributions as suggested by Emshoff and Sisson.

Student's T Test. Student's T test is a very powerful statistical test which can be used to compare the means of two populations. Such

a test might be used to indicate if the mean values of the simulation output over two adjacent time periods were statistically indistinguishable (Ref 25:343). This, in turn, would indicate that the simulation was in steady state. To use the test, however, one must be able to assume that the sample data is from at least a near normal or mound shaped probability distribution (Ref 12:169).

Since, in this study, the flying scenario calls for no flying on Saturday or Sunday, the distribution of manhours used per day is bimodal in nature. Consequently, the authors felt an assumption of normality was not warranted.

It is important to point out, nevertheless, that in other experiments having a continuous flying schedule, such as a wartime model or a peacetime scenario as described by Tetmeyer, the T test might be an appropriate tool (Ref 33:126-127). For a further discussion of the T test, the reader is referred to any elementary text on parametric statistics.

The Man-Whitney U Test. The Mann-Whitney U test is a nonparametric statistical test designed to indicate if two independent random samples can reasonably be expected to have come from different probability distributions, or if the two are stochastically indistinguishable. The primary advantage of this test is its applicability regardless of the probability distribution of the data to be tested (Ref 28:116).

Use of the U Test in Determining Steady State. In light of the distribution-free nature of the Mann-Whitney test, it was tested as an indicator of steady state on the simulation runs which had constrained

manpower. The first fourteen days of simulation output were compared to the remainder of the data from each run; a two week segment was chosen to provide sufficient data for meaningful application of the test, and also to avoid bias which would have resulted if proportionately more Saturdays and Sundays (nonflying days) were included in one of the samples than in the other. The null hypothesis (H_0) was defined to be:

H_0 : The two samples come from the same probability distribution.

If H_0 is not rejected, then the test is not able to distinguish between the two samples, and the authors considered steady state to exist at the beginning of the first sample. The alternate hypothesis (H_a) was defined as:

H_a : The two samples do not come from the same probability distribution.

When H_a is accepted, a significant difference exists between the two samples indicating that steady state has not been reached at the beginning of the first sample. In such instances, the authors threw out data from the first part of the time series (in increments of seven days) until the test indicated that steady state had been reached.

Application of the Mann-Whitney test involves first, computing the value of the U statistic for the two samples being compared. U is defined as the smaller of

$$U = N_1 N_2 \frac{N_1 + 1}{2} - W_1 \quad (4)$$

or
$$U = N_1 N_2 \frac{N_2(N_2 + 1)}{2} W_2 \quad (5)$$

where N_1 and N_2 are the sample sizes of the two time periods being compared, and W_1 and W_2 are obtained by ranking the $(N_1 + N_2)$ data points from one to $(N_1 + N_2)$ and summing these ranks for each group. Then, for N_1 and N_2 greater than ten.

$$Z = \frac{U - E(U)}{V(U)} \quad (6)$$

where Z is the standard normal random variable, $E(U)$ is the expected value of the U statistic

$$E(U) = \frac{N_1 N_2}{2} \quad (7)$$

and $V(U)$ is the variance of the U statistic.

$$V(U) = \frac{N_1 N_2 (N_1 + N_2 + 1)}{12} \quad (8)$$

(Ref 25:535-538).

If H_0 is true then

$$P(|Z| \leq Z_{\alpha/2}) = \alpha \quad (9)$$

where $\alpha/2$ is equal to the integral of the standard normal distribution function from $Z_{\alpha/2}$ to ∞ (Ref 25:276). It can then be said with probability $(1 - \alpha)$ that if the absolute value of Z , from equation (6), is greater than or equal to $Z_{\alpha/2}$ then H_0 is not true.

The Runs Test. The Runs Test was used to identify steady state for simulation runs with unconstrained manpower. This test is a distribution

free statistical test which can be used to guard against nonrandomness in a time series. Such nonrandomness may be related to either trends or periodicities in the data. As noted earlier in this paper, one criteria for steady state in a simulation is randomness of the output data about some central value. To use the test, the time series is separated into observations above the mean and observations below the mean. A run is defined as a series of observations all above or all below the mean. Given the number of data points above the mean (N_1), the number of points below the mean (N_2), and the number of runs (R); it is possible to compute the probability of observing R or fewer runs assuming the sequence is in fact random. These probabilities are commonly tabled and can be used in testing various hypotheses (Ref 25:542-547).

Application of Runs Test in Determining Steady State. In using the Runs Test as an indicator of steady state, the authors aggregated the daily manhour totals by weeks. Too few runs, in a statistical sense, were taken as an indication of a trend in the data. The null and alternate hypotheses were defined as follows:

H_0 : There is no trend in the data--steady state is assumed to exist over the entire time series.

H_a : There is a trend in the data--steady state does not exist over the entire time series.

The rejection region was defined to be R such that $P(R) \leq .1$. As with the Mann-Whitney Test, if H_0 was rejected with the original data set, the data set was reduced until H_0 was no longer rejected.

Type I and Type II Errors. When using any statistical test of hypothesis, the experimenter must be aware of the errors that can be made in accepting or rejecting H_0 . The probability of rejecting H_0 when H_0 is in fact true is termed the Type I error and is denoted by α . The authors chose the relatively large value of 0.1 for α , so that the Type II error (β), defined as the probability of accepting H_0 when actually H_a is true, would not be too large. Mendenhall and Shaeffer note that as α increases β will decrease (Ref 25:330). Even so, with the relatively small sample sizes involved in this study, β will be quite large. The authors, however, do not consider this serious since the intention is not to "prove" that the null hypothesis is true, but only to detect any significant deviation from that hypothesis.

Confidence Intervals

The mean value for manhours used, from the simulation output, is merely an estimate of the true mean for the system being modeled. The accuracy of this estimate is affected by the analyst's decision on when steady state exists, the number of data points used in the estimate (N), and the variance of the sample data (S^2). When considering the accuracy of such an estimate, the analyst is led to the consideration of statistical confidence intervals.

If the sample data is composed of independent random variables, the sample mean will possess a normal distribution for sample sizes larger than thirty (Ref 25:270). Then, the statistic

$$Z = \frac{(\bar{X} - \mu)}{\sqrt{S^2/n}} \quad (10)$$

where μ = the population mean

\bar{X} = the sample mean

S_{μ}^2 = the sample variance of the mean ($\frac{S^2}{N}$)

S_{μ} = the sample standard deviation of the mean ($\sqrt{S_{\mu}^2}$)

As before, Z is the standard normal random variable with zero mean and variance equal to one. From equation (10) can be constructed the following confidence interval:

$$\mu = \bar{X} \pm Z_{\alpha/2} S_{\mu} \quad (11)$$

where $Z_{\alpha/2}$ is such that the integral of the standard normal density function from $Z_{\alpha/2}$ to ∞ is equal to $\alpha/2$. It can now be stated with probability $(1 - \alpha)$ that the true population mean is contained in the interval given in equation (11) (Ref 25:276-277).

Autocorrelation. In many simulation models, the parameter values for separate observations are not independent; they are affected in some way by those values that precede them. For instance, in the LCOM model used in this study, high work loads on one day may carry over into the next day causing the man hours used on both days to be high. This type of dependence is called autocorrelation. The autocorrelation coefficient, ρ , where ρ is as defined by equation (12), is a measure of this dependency. If man hours used on any particular day is assumed to be normally distributed, ρ equal zero implies independence, and ρ equal one implies complete dependence (Ref 35:70).

$$\rho(L) = \frac{E([X_t - \bar{X}][X_{t+L} - \bar{X}])}{\sigma^2} \quad (12)$$

where E = expected value

x_t = output value at time (t)

\bar{x} = sample mean

σ^2 = population variance

L = interval or lag between data points to be tested for dependence (Ref 10:194-199).

Given only a sample from the entire population, it is not possible to compute $\rho(L)$. However, an estimate, $\hat{\rho}(L)$ is available as follows.

$$\hat{\rho}(L) = \frac{\sum_{t=1}^{N-L} (x_t - \bar{x})(x_{t+L} - \bar{x})}{N \sum_{t=1}^{N-L} (x_t - \bar{x})^2} \quad (13)$$

Correcting for Autocorrelation. The procedure developed previously for constructing confidence intervals employed an estimate of the variance of the mean, S_{μ}^2 , based on an assumption of independence. If autocorrelation is present, then, the procedure for constructing confidence intervals must be modified.

One possibility is to examine $\hat{\rho}(L)$. If a value of (L) can be found such that $\hat{\rho}(L) = 0$, then the data can be combined or blocked into intervals of length (L). The mean values for these blocks will now be independent, and as such can be used to construct confidence intervals as previously described (Ref 10:199-200).

A second possibility is to correct the variance of the mean (S_{μ}^2) by using the following relationship.

$$\text{Corrected } S_{\mu}^2 = S_{\mu}^2 \left[1 + 2 \sum_{L=1}^{N-1} \left(1 - \frac{L}{N} \right) \hat{\rho}(L) \right] \quad (14)$$

The reader should note that when $\hat{\rho}(L) = 0$, the above reduces to the form used in the previous development of confidence intervals (Ref 10:199).

Sensitivity Analysis

The Moody Regression Program, discussed in Chapter II, was used to develop a relationship between direct manning requirements and flying hours per month. A regression equation was developed for each AFSC with constrained manpower only, constrained manpower and parts, and constrained manpower, parts and ATS. These equations were then plotted to demonstrate the sensitivity of manning requirements, for each AFSC, to various sortie rates and parts/ATS constraints.

The primary inputs to the Regression Program consist of flying hours per month and final direct manning (M from equation (3)) rounded in accordance with AFM 25-5 (Ref 2:7-8). Manning for some LCOM shred-outs is combined under a single AFSC as shown in Appendix A. Further, direct manning for some LCOM shred-outs cannot be properly estimated using LCOM hours from the PSR. For instance, the End of Runway Crew requires very few LCOM hours in the model. In reality though, the positions must be manned during all flying operations. Direct manning in this case was estimated using the following relationship (Ref 3:2-4).

$$M = \frac{(\text{crew size})(\text{work days/wk})(\text{hrs/day})(4.348 \text{ wks/mo})}{144 \text{ hours/man/month}} \quad (15)$$

Direct manning for these LCOM shred-outs was combined under the appropriate AFSC in the Regression Program.

Developing a Manning Document

The final step in obtaining an estimate of required manpower involves the use of the Moody Manpower Program. Up until this point only

direct manning has been discussed; that is, no consideration has been given to the fact that there is a requirement for supervisory and other overhead personnel in the maintenance organization being modeled. The Moody Manpower Program converts the results of the final simulation runs and the Moody Regression runs into a completed manning document, which reflects the total maintenance manning, by AFSC and work center, necessary to support the organization under study at a specified sortie rate and with specified constraints on parts/ATS.

In this study the authors used the Moody Manpower Program to develop estimates for total required manning, and also to demonstrate the sensitivity of total manning to parts/ATS constraints.

Summary

This chapter has included a discussion of the concepts and procedures used by the authors in accomplishing the objectives of this study. In the next chapter the authors demonstrate the methodology discussed here and present the results.

V. ANALYSIS AND RESULTS

The analysis and results of the F-15 peacetime LCOM study are documented in this chapter. The analysis section investigates the simulation's steady state conditions and autocorrelation functions. The results section consists of maintenance manpower estimates for each AFSC listed in Table I of Chapter III, statistical confidence intervals for these estimates, manpower sensitivity to variations in spare parts and ATS, and a complete basic manning document for the modeled F-15 TFW.

Analysis

Simulation analysis consisted of graphical and statistical interpretation of accomplished aircraft sortie rate, AFSC manhours used, and AFSC autocorrelation coefficients. The authors used the procedures discussed in Chapter IV to analyze these variables. The simulation run length during this analysis consisted of 98 simulation days including weekends; this period is equivalent to 14 seven day weeks.

Steady State. The authors conducted steady state analysis on both unconstrained and constrained manpower simulations. During the unconstrained manpower simulations, PSR's were requested every seven days. The authors used daily PSR's for the constrained manpower simulations.

The authors chose a seven day data period for unconstrained manpower simulations due to a fundamental seven day cycle in the LCOM statistical output. This cycle was caused primarily by the peacetime operational scenarios; the five day flying and seven day maintenance schedule resulted in an appreciable decrease in manhours used for most AFSC's during the weekends. The accumulation of output data into seven day intervals smoothed this cyclic pattern and simplified the steady state

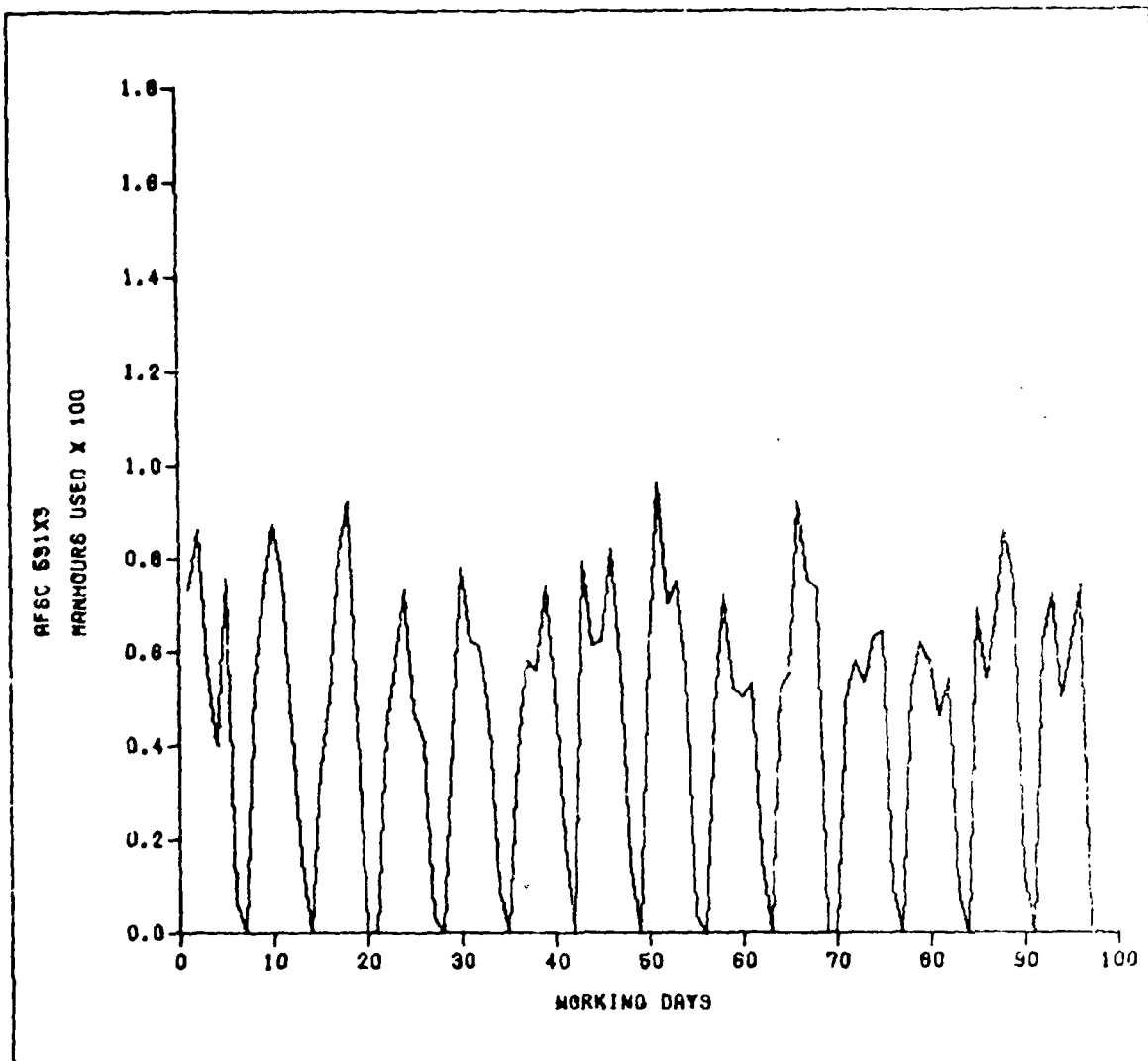


Figure 10. AFSC 531X3 (Structural Repair) Daily Manhours Used

analysis. For example, Figure 10 graphically depicts the daily manhours used for AFSC 531X3 (Structural Repair) over the 93 day simulation period based on unconstrained parts/ATS at a .74 scheduled sortie rate. The graph illustrates the seven day cyclic pattern caused by decreased man-hours used during weekends. Figure 11 contains the weekly manhours used for the same AFSC with identical simulation conditions. The graph, in this case, clearly represents steady state conditions throughout the

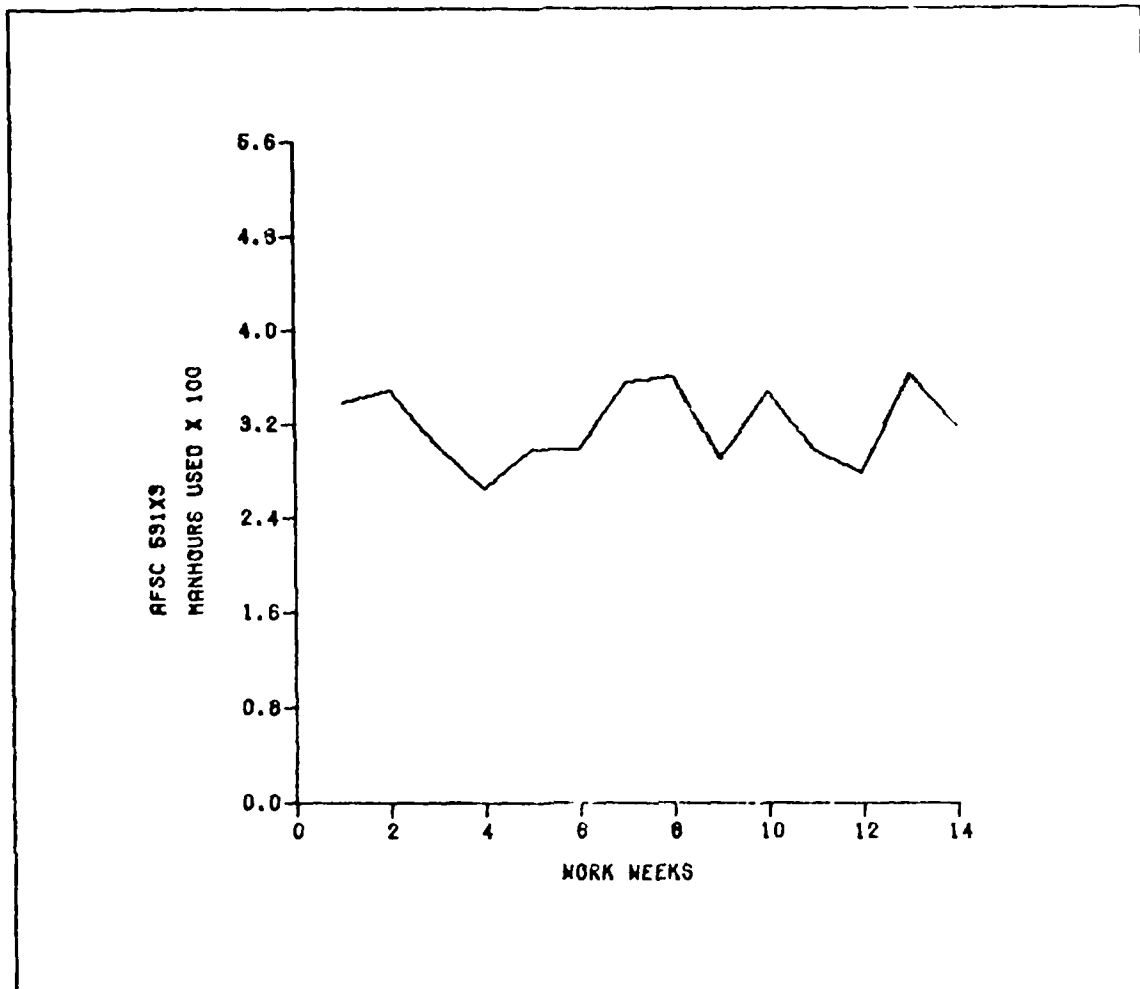


Figure 11. AFSC 531X3 (Structural Repair) Weekly Manhours Used

14 week period; additionally, the Runs test described in Chapter IV indicated no trends in this weekly data. All AFSC's except AFSC 423X3 (Fuel Systems) exhibited similar graphical and statistical steady state conditions throughout the 98 day unconstrained manpower simulations.

AFSC 423X3 was the only manpower statistic to exhibit nonsteady state conditions during the unconstrained manpower simulations. This AFSC exhibited an unstable condition during the constrained parts/ATS simulation at a .43 scheduled sortie rate. Figure 12 graphically

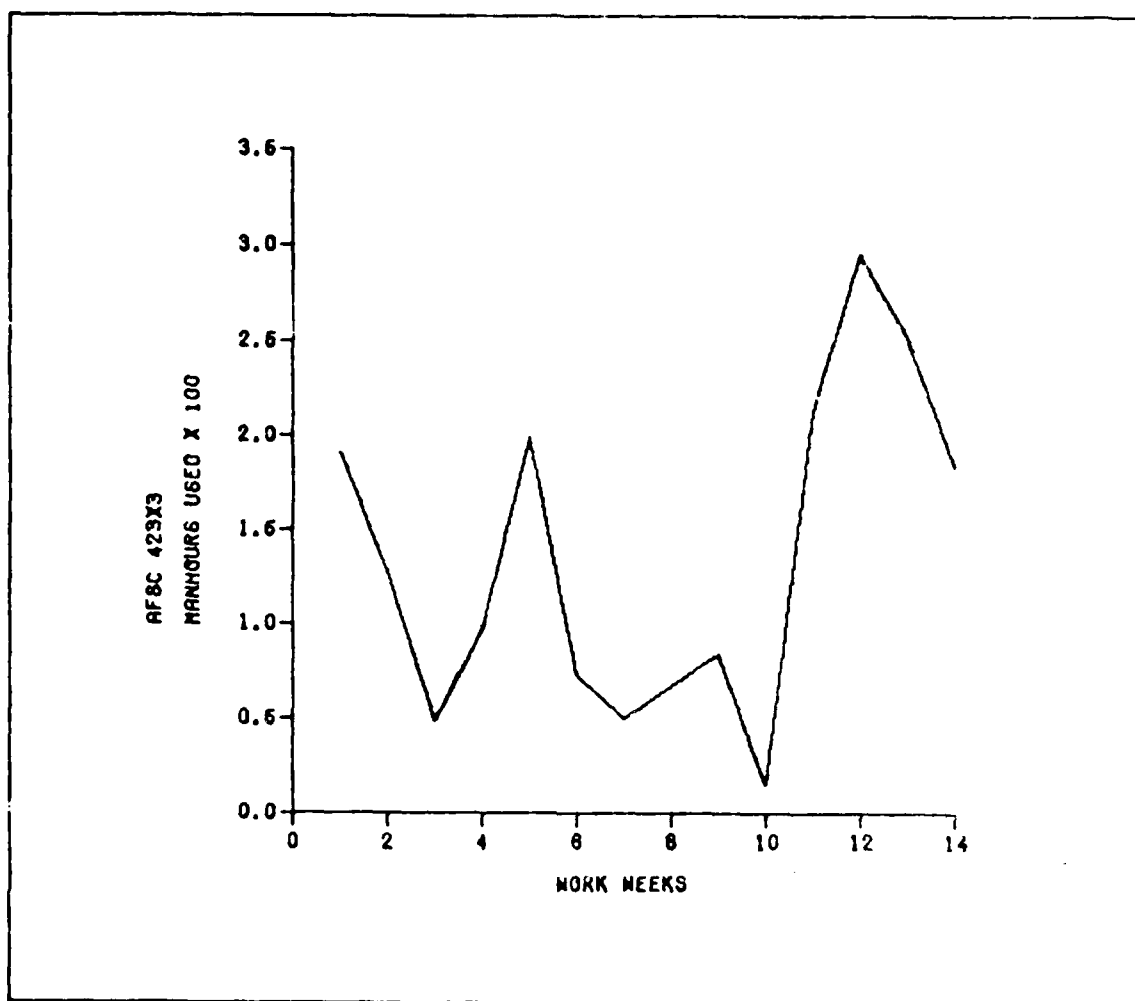


Figure 12. AFSC 423X3 (Fuel Systems) Weekly Manhours Used

illustrates this AFSC's instability in manhours used. The increase in manhours used during the eleventh through thirteenth weeks is especially noteworthy. This demand indicates a backlog of work during the latter portion of the simulation period. The authors, therefore, constrained manpower for this AFSC using the average daily manhours used during the eleventh through thirteenth weeks. This method of constraining manpower has two advantages: first, manpower is allocated to the constrained manpower simulation based on peak workloads as described by Tetmeyer (Ref 33:119); second, if peak workloads are not considered during the

manpower constraint process, the constrained manpower simulation's PSR will indicate manhour utilization for the AFSC in question greater than 60 percent and an additional constrained manpower simulation will be necessary to reduce this utilization below 60 percent.

Table III
AFSC 531X3 Manpower Constraint Calculation

Source	Calculation
Equation (2)	$M_s = \frac{\text{(0-98 Day PSR Manhours Used)}}{\text{(Utilization Factor)} \text{(Number of Days)} \text{(Shift Length)}}$ $= \frac{4445}{(.6)(98)(8)}$ $= 9.449 \approx 10 \text{ Men}$

Since all other AFSC's exhibited steady state conditions during the entire simulation period, the authors constrained manpower for these AFSC's using equation (2) from Chapter IV and the respective AFSC man-hours used contained in the 0-98 day cumulative PSR. Table III illustrates the calculation of the manpower constraint for AFSC 531X3 (Structural Repair) using equation (2) and the respective manhours used in Figure C-1 of Appendix C. It should be noted that all manpower constraints are rounded to the next highest integer value during input to the main LCOM program.

During the constrained manpower simulations, the authors used a modified version of LCOM (Ref:19) which provided daily PSR's containing

only operations and personnel data. The elimination of shop repair and supply data from the PSR's appreciably reduced the amount of unneeded computer output during these simulations. Daily manhours used data was required in order to accurately compute the autocorrelation coefficients and statistical confidence intervals discussed later in the chapter; the daily data permits large sample size statistical analysis and increases the statistical confidence in the output data.

Figure 13 graphically illustrates the daily manhours used for AFSC 431X1 (Flight Line Crew Chief) for the unconstrained parts/ATS simulation at a .74 scheduled sortie rate. This graph indicates two major points: first, the seven day cycle of manhours used is very evident; second, the underlying steady state nature of the data throughout the 98 day simulation period is visible. Because steady state conditions are more difficult to detect in the daily manhours used data, the Mann-Whitney Test described in Chapter IV was used to verify output stability. The test indicated no statistical differences in the data contained in Figure 13.

All AFSC's except those listed in Table IV exhibited similar steady state conditions during the entire 98 day constrained manpower simulation period. For those AFSC's in Table IV, the Mann-Whitney test initially indicated instability in the data. The test was, therefore, reapplied to these AFSC's after eliminating the first seven days of data. In most cases, the reduced data set passed the test. In other cases, the first 14 days of data had to be eliminated before the reduced data set would pass the Mann-Whitney test. Table IV indicates each AFSC and its respective usable data sub-set. These data subsets were used to verify

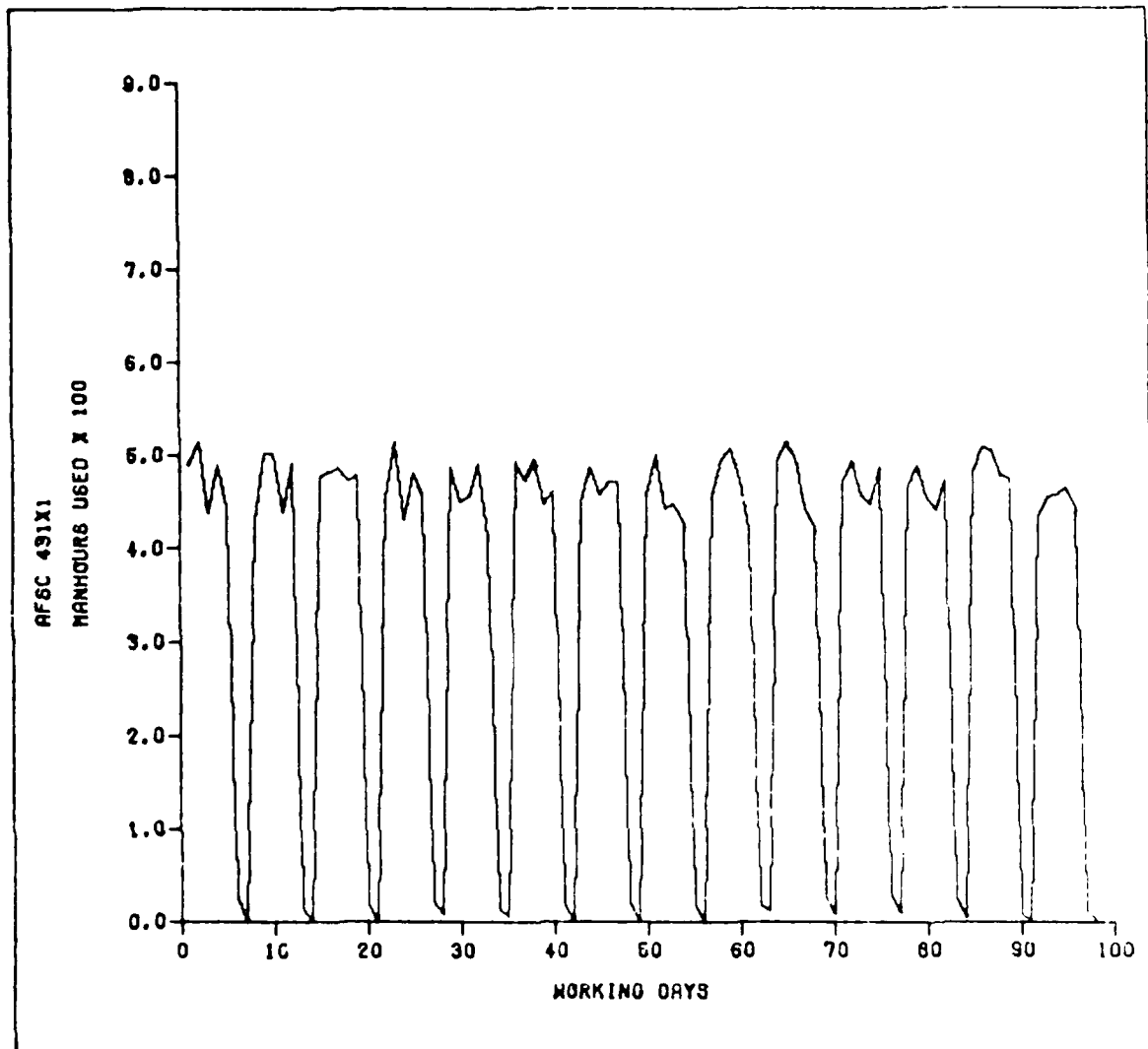


Figure 13. AFSC 431X1 (Flight Line Crew Chief) Daily Manhours used

manhour utilization, calculate manpower estimates, and compute confidence intervals for the AFSC's listed in Table IV.

The authors also analyzed accomplished aircraft sortie rate for the constrained manpower simulations. If this sortie rate exhibits steady state conditions, then the modeled F-15 TFTW can satisfy TAC training syllabi requirements on a routine basis.

Table IV

LCOM AFSC's That Exhibited Initial Transient Conditions

Constraint Type	Scheduled Sortie Rate	AFSC	Usable Simulation Period
Unconstrained Parts/ATS	1.0	326X2A (COM/ NAV/ ECM)	Last 91 Days
Constrained Parts/ Unconstrained ATS	1.0	423X3 (Fuel Systems)	Last 84 Days
Constrained Parts/ Constrained ATS	1.0	326X1D (Automatic Test Station)	Last 91 Days
	1.0	326X2A (Inertial Navigation System)	Last 91 Days
	1.0	431X1 (Phase Inspection)	Last 91 Days
	.74	426X2 (Jet Engine)	Last 84 Days

Figure 14 depicts the accomplished aircraft sortie rate for the unconstrained parts/ATS simulation at a .74 scheduled sortie rate. The graph excludes weekends due to the undefined nature of sortie rate for nonflying days. In Figure 14, accomplished sortie rate exhibits a steady state condition throughout the 70 flying days. The Mann-Whitney test also verified steady state conditions in the data. Accomplished sortie rates

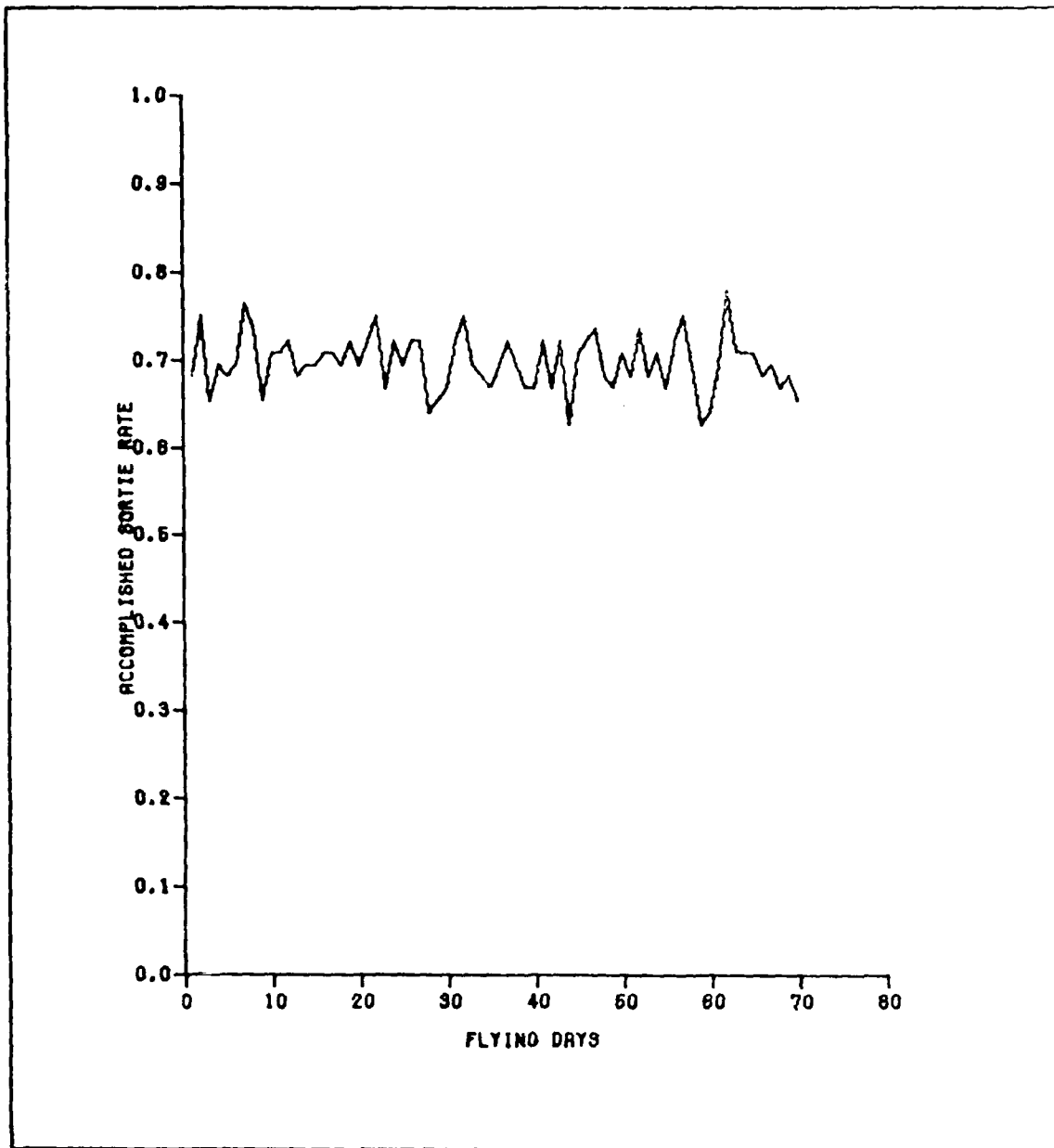


Figure 14. Accomplished Aircraft Sortie Rate for Constrained Manpower and Unconstrained Parts/ATS Simulation at .74 Scheduled Sortie Rate

for each of the nine combinations of parts/ATS constraints and scheduled sortie rates exhibited similar steady state conditions. These steady state conditions indicate that the modeled F-15 TFTW can routinely meet syllabi training requirements.

Autocorrelation. The authors conducted the autocorrelation analysis on the constrained manpower simulations. The autocorrelation coefficients ($\hat{\rho}(L)$) were estimated for each AFSC using the respective manhours used and equation (13) found in Chapter IV; the lag (L) between daily observations was incremented to include a one through 97 day lag. For those AFSC's listed in Table IV, $\hat{\rho}(L)$ was estimated for the corresponding reduced data sets. The analysis consisted of both graphical and statistical interpretation of the autocorrelation function.

Figure 15 graphically illustrates the autocorrelation function versus the lag in days for AFSC 431X1 (Flight Line Crew Chief) for the unconstrained parts/ATS simulation at a .74 scheduled sortie rate. This graph indicates two major points: first, the seven day cyclic nature of the function caused by the reduced manhour demands on weekends is clearly evident; second, multiple lags of seven days are highly correlated. The high correlation of manhours used on multiple seven day lags is due to the seven day cyclic nature of the simulation data. This functional form of autocorrelation coefficients prevents the grouping of manhours used into blocks of independent observations. The authors, therefore chose to use the estimated correlation coefficients and equation (14) to correct the variance of daily manhour observations during the computation of manpower statistical confidence intervals.

AFSC 423X3 (Fuel Systems) was the only AFSC to exhibit a more desirable autocorrelation function. The reason for this unique difference was that the fuel systems shop did not show a routine decrease in manhours used each weekend. This resulted in an absence of the seven day cycle. Figure 16 illustrates the graphical nature of the autocorrelation function for this AFSC. In the graph, lags of 15 or more days indicate

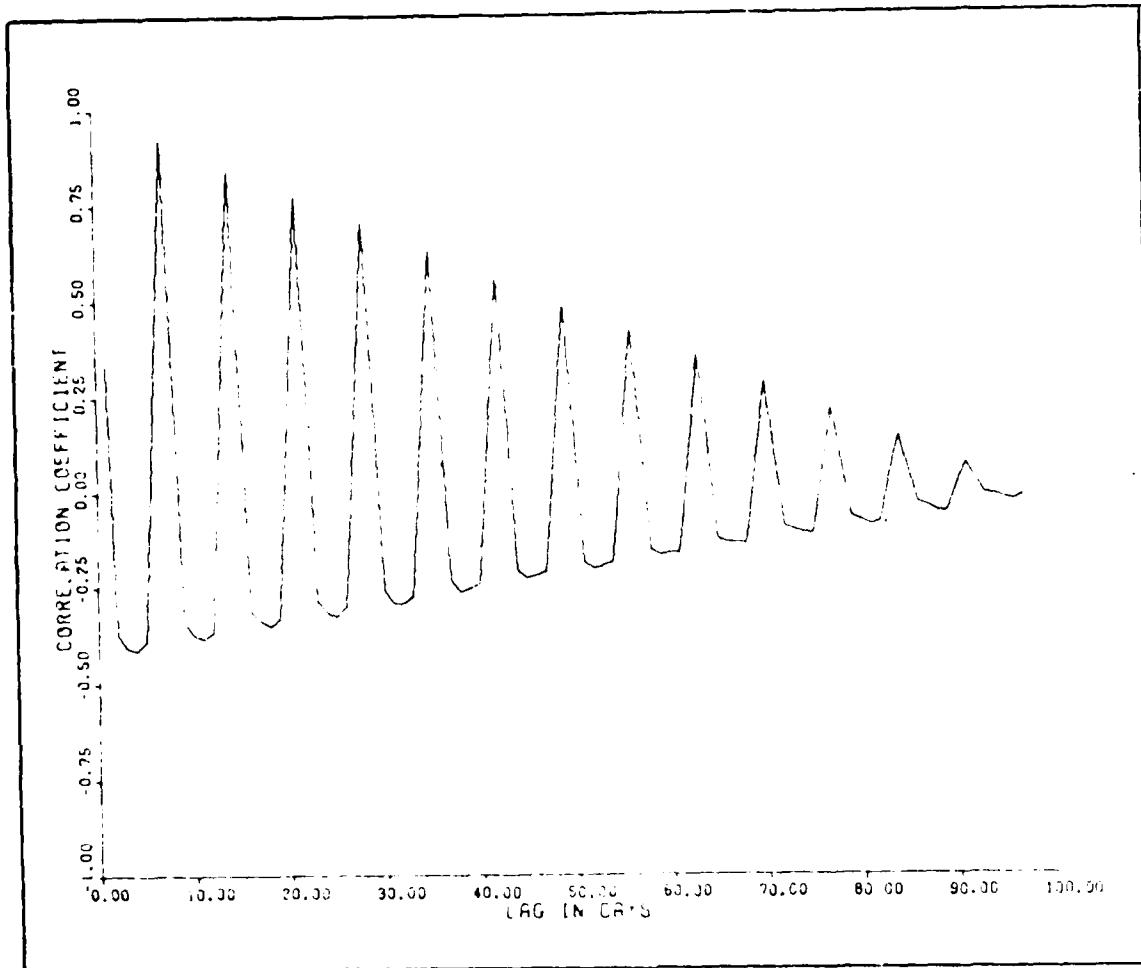


Figure 15. AFSC 431X1 (Flight Line Crew Chief) Autocorrelation Function
 a $\hat{\rho}(L)$ close to zero. As a result, manhours used contained on cumulative 15 day PSR's would be statistically independent and could be treated as independent observations during the calculation of statistical confidence intervals. The type of autocorrelation function illustrated in Figure 16 is often found in an LCOM simulation where maintenance and flying activity operate concurrently. Concurrent maintenance and flying activity do not produce the cyclic output data that is found in this study. In particular, wartime LCOM simulations contain the functional form illustrated in Figure 16.

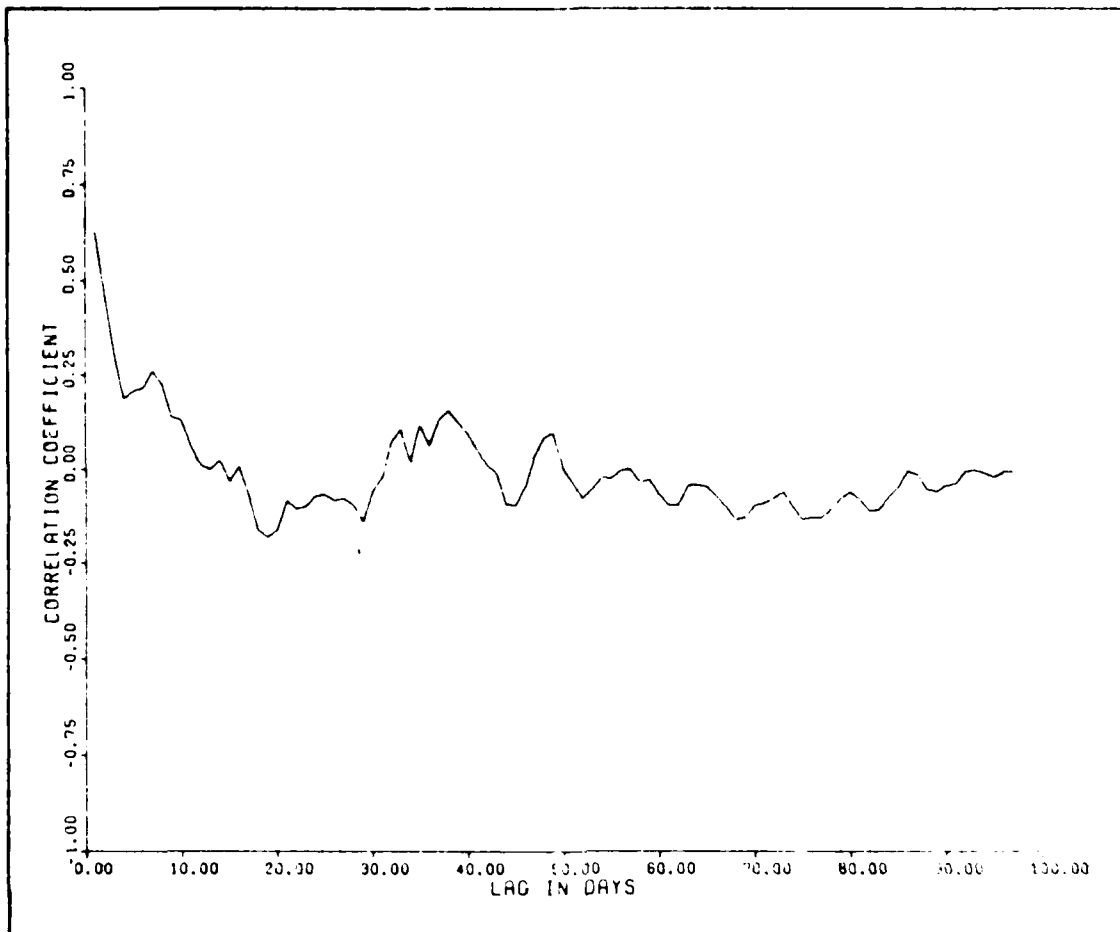


Figure 16. AFSC 423X3 (Fuel Systems) Autocorrelation Function

Because the majority of AFSC's in this study exhibited autocorrelation functions similar to Figure 15, the variance of daily manhours used for each AFSC including AFSC 423X3 was corrected using equation (13). These corrected variances were used during calculation of the manpower confidence intervals.

Results

The results of the F-15 TFTW LCOM simulation contain the direct manning estimations (M) for each AFSC contained in Table I of Chapter III,

95 percent confidence intervals for these estimates, and sensitivity of M to constrained parts and constrained parts/ATS. The final output of these estimates is a USAF basic manning document for the F-15 TFTW based on unconstrained parts and ATS.

Direct Manning. The average daily manhours used from the final constrained manpower simulations are applied to equations (1) and (2) from Chapter IV to calculate the direct manning for each AFSC listed in Table I. Direct Manning is then converted to whole men according to AFM 25-5 standards (Ref 2:Chap. 6, p. 29). During the calculation of M, the AFM 25-5 standard of 30.44 work days per month is applied to equation (2). Table V illustrates the computation of direct manning for AFSC 531X3 (Structural Repair) for the unconstrained parts/ATS simulation at .74 scheduled sortie rate.

Table V
AFSC 531X3 Direct Manning Computation

Source	Calculation
Equation (1)	$M_s = \frac{\text{Average Daily Manhours Used}}{(\text{Utilization Factor})(\text{Shift Length})}$ $= \frac{(40.44)}{(.6)(8)}$ $= 8.425$
Equation (2)	$M = \frac{M_s (\text{Workdays/mo})(\text{Shift Length})}{144 \text{ Hours/Man/Month}}$ $= \frac{(8.425)(30.44)(8)}{144}$ $= 14.248$
AFM 25-5	Direct Manning = 14 Men

Confidence Intervals. Confidence intervals for direct manning are computed using equations (1), (2), (11), and (14) found in Chapter IV. The computation is a six step process: first, estimate the sample variance of daily manhours used (Ref 25:268); second, use equation (14) to calculate the corrected sample variance of daily manhours used; third, convert the corrected sample variance into a standard deviation of daily manhours used; fourth use equations (1) and (2) to transform the standard deviation of daily manhours used into a direct manning standard deviation; fifth, use equation (11) to develop 95 percent confidence intervals for direct manning; finally, convert the confidence intervals into integer values in accordance with AFM 25-5 (Ref 2: Chap. 6, p. 29). Table VI illustrates the computation of a 95 percent confidence interval for the direct manning of AFSC 531X3 based on unconstrained parts/ATS at a .74 scheduled sortie rate. In these calculations, N equals the number of steady state days simulated. In all cases, except those AFSC's listed in Table IV, N equals 98.

The procedures contained in Tables V and VI were used to compute each AFSC's direct manning and confidence intervals for the three constraint types (unconstrained parts/ATS, constrained parts/unconstrained ATS, and constrained parts/ATS) and scheduled sortie rates (.43, .74, and 1.0). The direct manning for the End of Runway Crew was separately calculated using equation (15) found in Chapter IV and added to the direct manning estimate and confidence interval of AFSC 431X1 (Flight Line Crew Chief). This calculation is illustrated in Table VII.

Table VI
AFSC 531X3 Confidence Interval Computation

Source	Calculation
Sample Variance of Daily Manhours Used	$S_{\mu}^2 = 5.526$
Equation (14)	$\text{Corrected } S_{\mu}^2 = S_{\mu}^2 \left[1 + 2 \sum_{L=1}^{N-1} \left(1 - \frac{L}{N}\right) \hat{p}(L) \right]$ $= 5.256 \left[1 + 2 \sum_{L=1}^{97} \left(1 - \frac{L}{98}\right) \hat{p}(L) \right]$ $= .507$
Convert Variance to Standard Deviation	<p>Average Daily Manhours Used = $\sqrt{.507}$</p> <p>Standard Deviation = .712</p>
Equation (1)	$M_S \text{ Standard Deviation} = \frac{\text{(Average Daily Manhours Used Standard Deviation)}}{\text{(Utilization Factor)} \text{(Shift Length)}}$ $= \frac{(.712)}{(.6)(8)}$ $= .148$
Equation (2)	$M \text{ Standard Deviation} = \frac{\text{(} M_S \text{ Standard Deviation)} \text{(Work Days/Mo)} \text{(Shift Length)}}{\text{(144 Hours/Man/Month)}}$ $= \frac{(.148)(30.44)(8)}{144}$ $= .250$

Table VI. (Continued)

AFSC 531X3 Confidence Interval Computation

Source	Calculation
Equation (11)	$95\% \text{ Confidence Interval} = M \pm (Z_{\alpha/2}) (M \text{ Standard Deviation})$ $= 14.248 \pm (1.96)(.250)$ $= 14.248 \pm .49$ $= (13.758, 14.738)$
AFM 25-5	$95\% \text{ Confidence Interval for } M = (13, 14) \text{ Men}$

Table VII

End of Runway Crew Direct Manning Computation

Source	Computation
Equation (15)	$M = \frac{(\text{Crew Size})(\text{Workdays per Month})(\text{Hours per Day})(4.348 \text{ Weeks per Month})}{144 \text{ Hours/Man/Month}}$ $= \frac{(3)(5)(16)(4.348)}{144}$ $= 7.25$
AFM 25-5	Direct Manning = 7 Men

Direct Manning Sensitivity. This section illustrates both graphically and statistically the sensitivity of direct manning estimates to each of the three simulation constraint types and scheduled sortie rates. The authors used the Moody Regression Program to develop a regression equation for each constraint type comparing AFSC direct manning with flying hours per month (FHPM). The accomplished sortie rate (ASR) corresponding to the simulation's scheduled sortie rate was converted into FHPM for the purpose of this comparison. Table VIII depicts the resulting ASR and FHPM for the three constraint types and scheduled sortie rates.

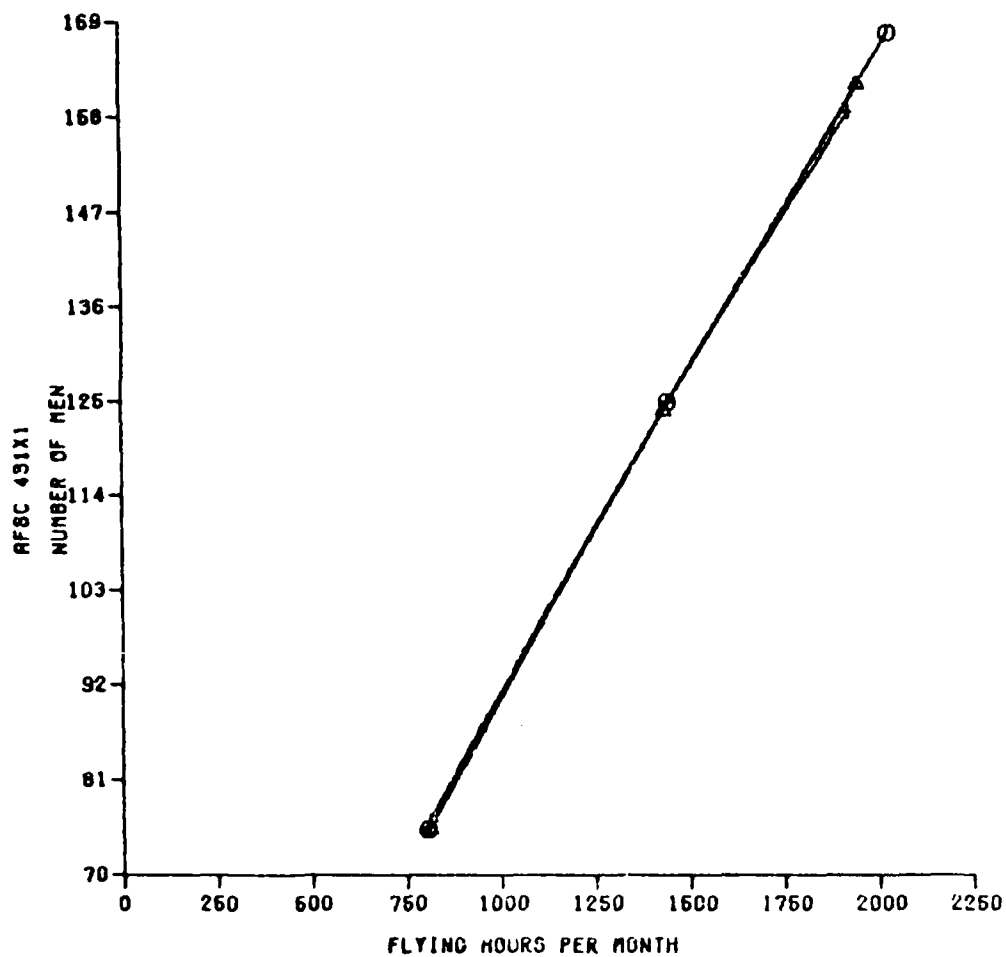
Table VIII

Accomplished Sortie Rate (ASR) and Flying Hours per Month (FHPM) versus Constraint Type and Scheduled Sortie Rate

Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
Unconstrained Parts/ Unconstrained ATS	.404 ASR 806 FHPM	.697 ASR 1444 FHPM	.955 ASR 2037 FHPM
Constrained Parts/ Unconstrained ATS	.406 ASR 809 FHPM	.692 ASR 1433 FHPM	.916 ASR 1954 FHPM
Constrained Parts/ Constrained ATS	.406 ASR 808 FHPM	.698 ASR 1446 FHPM	.904 ASR 1930 FHPM

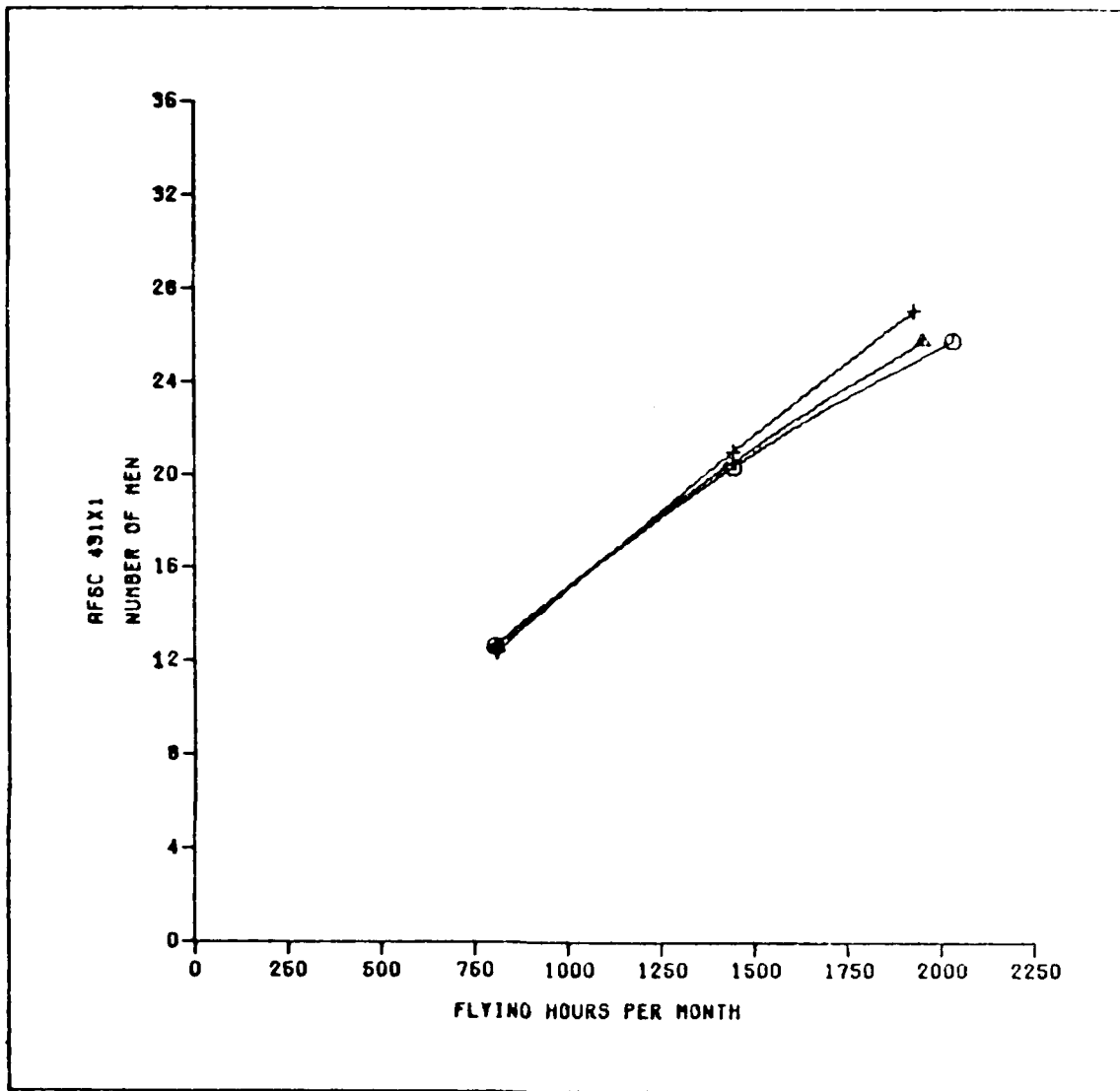
Figures 17 through 30 depict the direct manning for those AFSC's listed in Table I. These figures use the Moody Regression equations to graphically illustrate the sensitivity of direct manning to constraint type and flying hours per month. The figures also depict in tabular form the direct manning and 95 percent confidence intervals corresponding to each constraint type and scheduled sortie rate.

Table VIII and Figures 17 through 30 summarize the relationships between scheduled sortie rate, flying hours per month, accomplished sortie rate, and direct manning. For example, AFSC 431X1 (Flight Line Crew Chief) in Figure 17 requires 76 direct men with a 95 percent confidence interval of (74-78) to achieve 806 FHPM and a .404 ASR based on unconstrained parts/ATS and a .43 scheduled sortie rate.



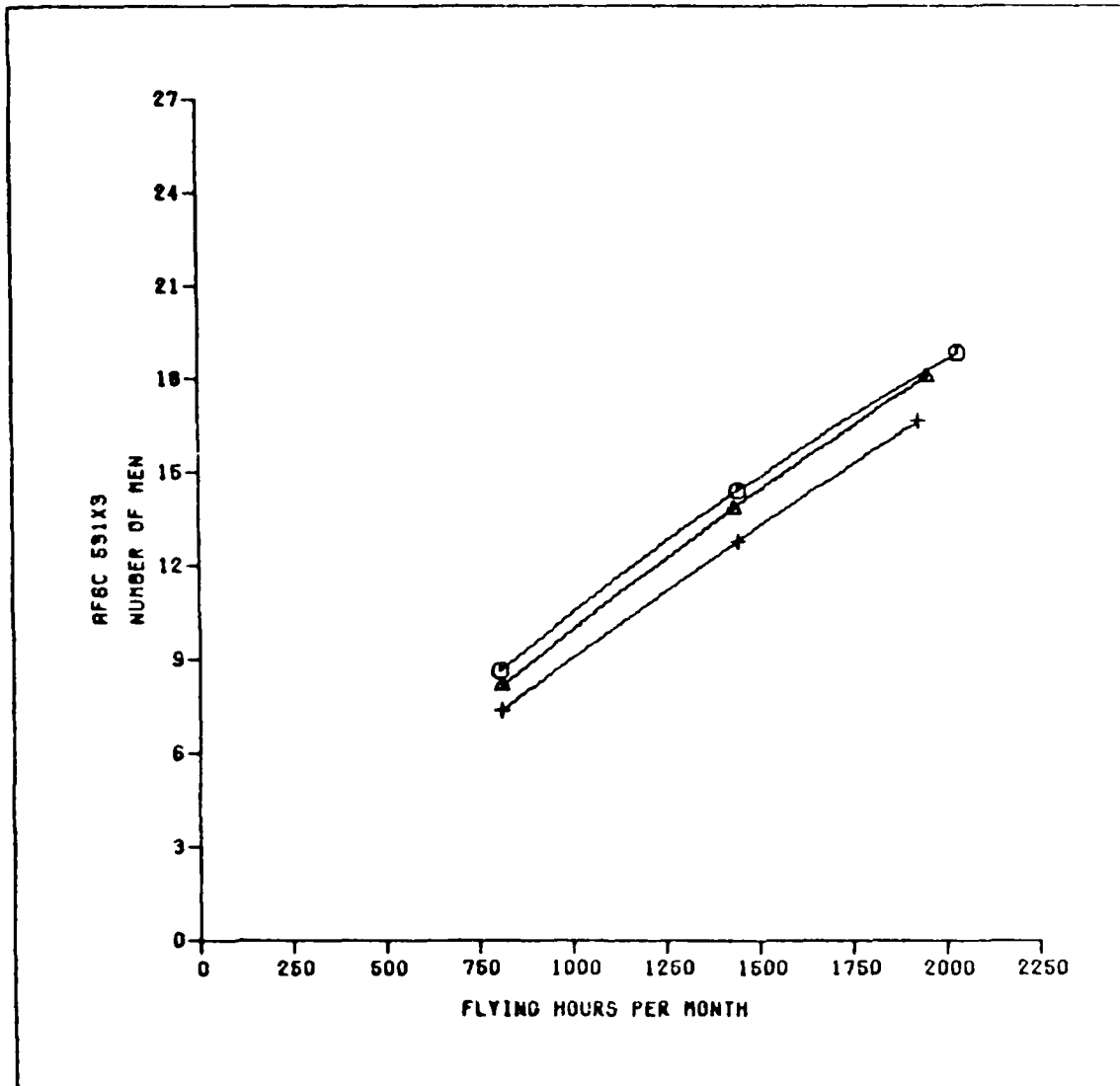
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
0 - Unconstrained Parts/ Unconstrained ATS	76 (74-78)	124 (120-128)	168 (163-172)
△ - Constrained Parts/ Unconstrained ATS	76 (74-79)	123 (119-127)	162 (157-167)
+ - Constrained Parts/ Constrained ATS	77 (75-79)	124 (119-128)	159 (149-169)

Figure 17. AFSC 431X1 (Flight Line Crew Chief) Direct Manning



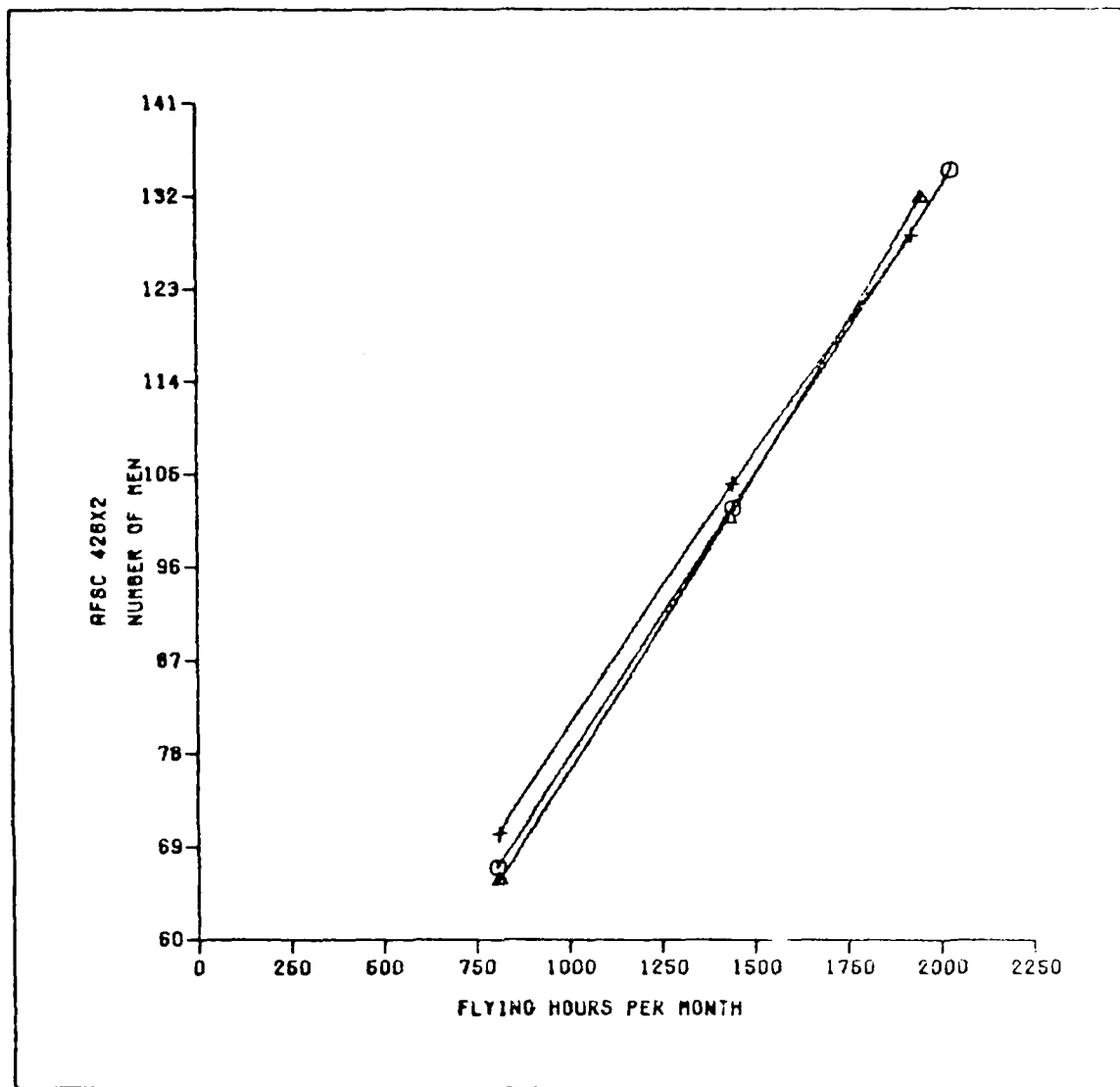
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
O - Unconstrained Parts/ Unconstrained ATS	13 (12-13)	20 (20-21)	26 (26-27)
Δ - Constrained Parts/ Unconstrained ATS	13 (12-13)	20 (19-22)	26 (25-27)
+ - Constrained Parts/ Constrained ATS	14 (13-14)	19 (17-20)	28 (26-29)

Figure 18. AFSC 431X1 (Phase Inspection) Direct Manning



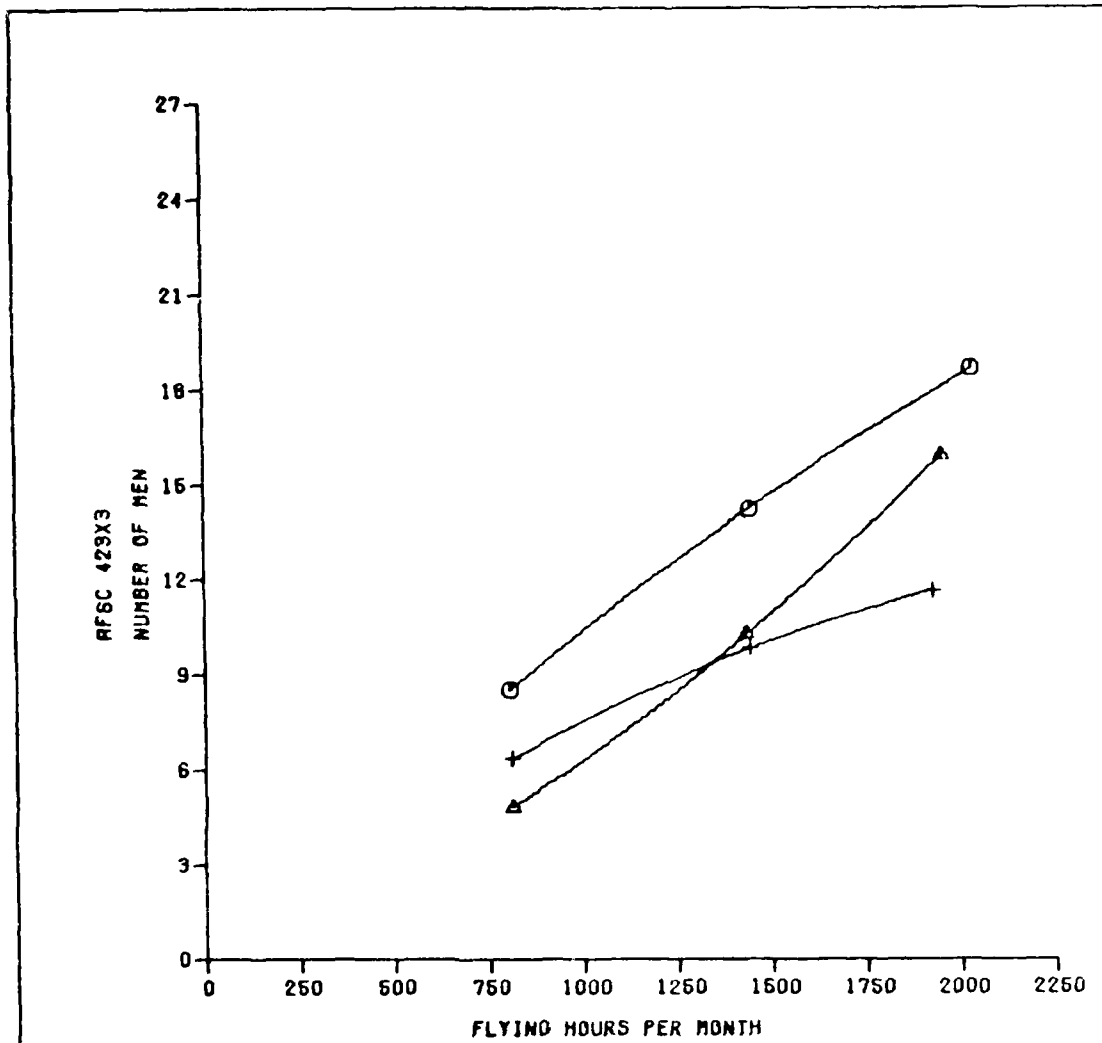
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
0 - Unconstrained Parts/ Unconstrained ATS	9 (8-9)	14 (13-14)	19 (18-20)
Δ - Constrained Parts/ Unconstrained ATS	8 (8-9)	14 (13-14)	18 (17-19)
+ - Constrained Parts/ Constrained ATS	8 (8-9)	12 (12-13)	17 (16-19)

Figure 19. AFSC 531X3 (Structural Repair) Direct Manning



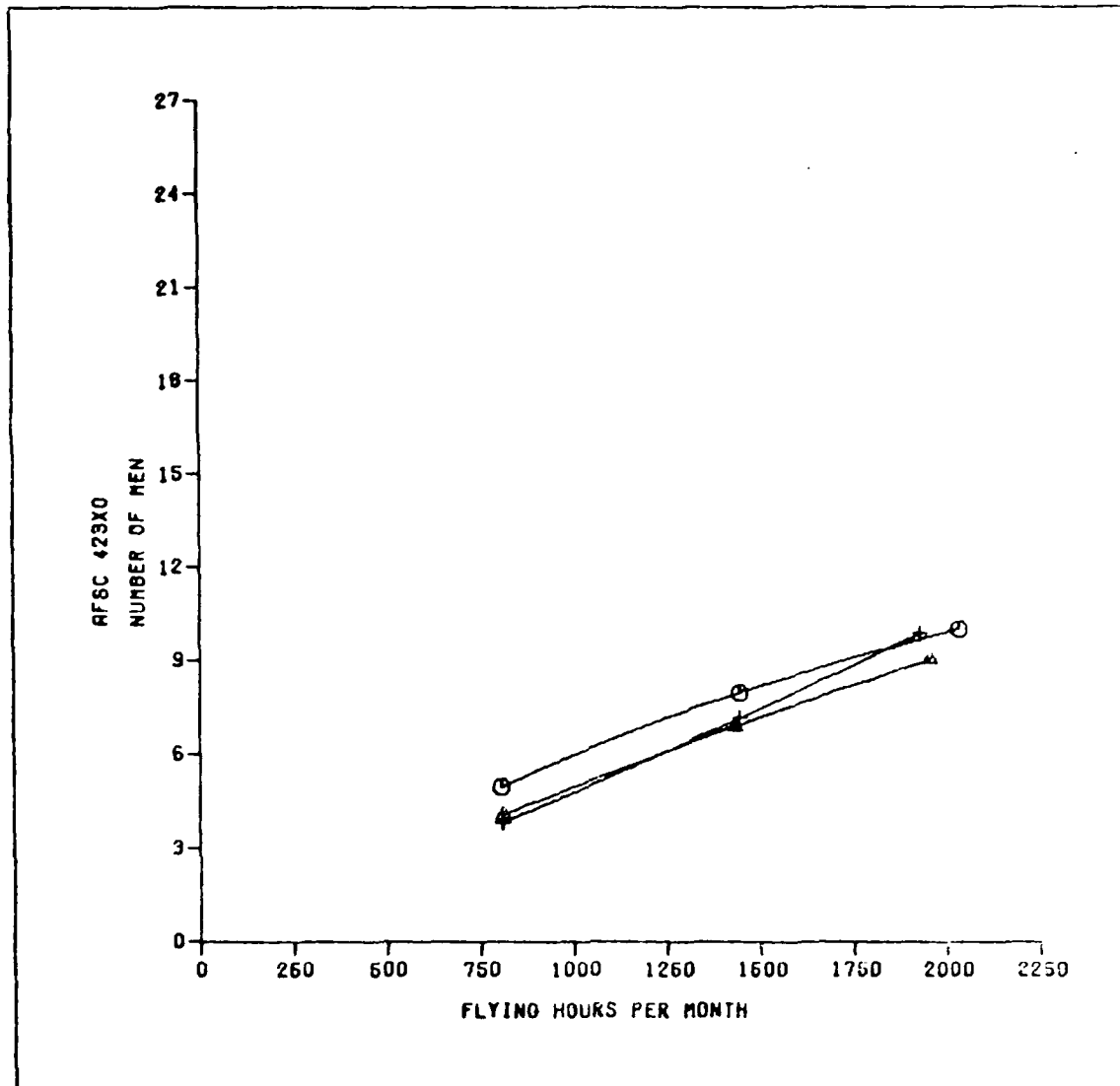
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
O - Unconstrained Parts/ Unconstrained ATS	66 (64-68)	103 (94-112)	134 (128-141)
Δ - Constrained Parts/ Unconstrained ATS	68 (65-70)	98 (92-107)	135 (128-139)
+ - Constrained Parts/ Constrained ATS	68 (65-72)	107 (101-112)	137 (117-147)

Figure 20. AFSC 426X2 (Jet Engine) Direct Manning



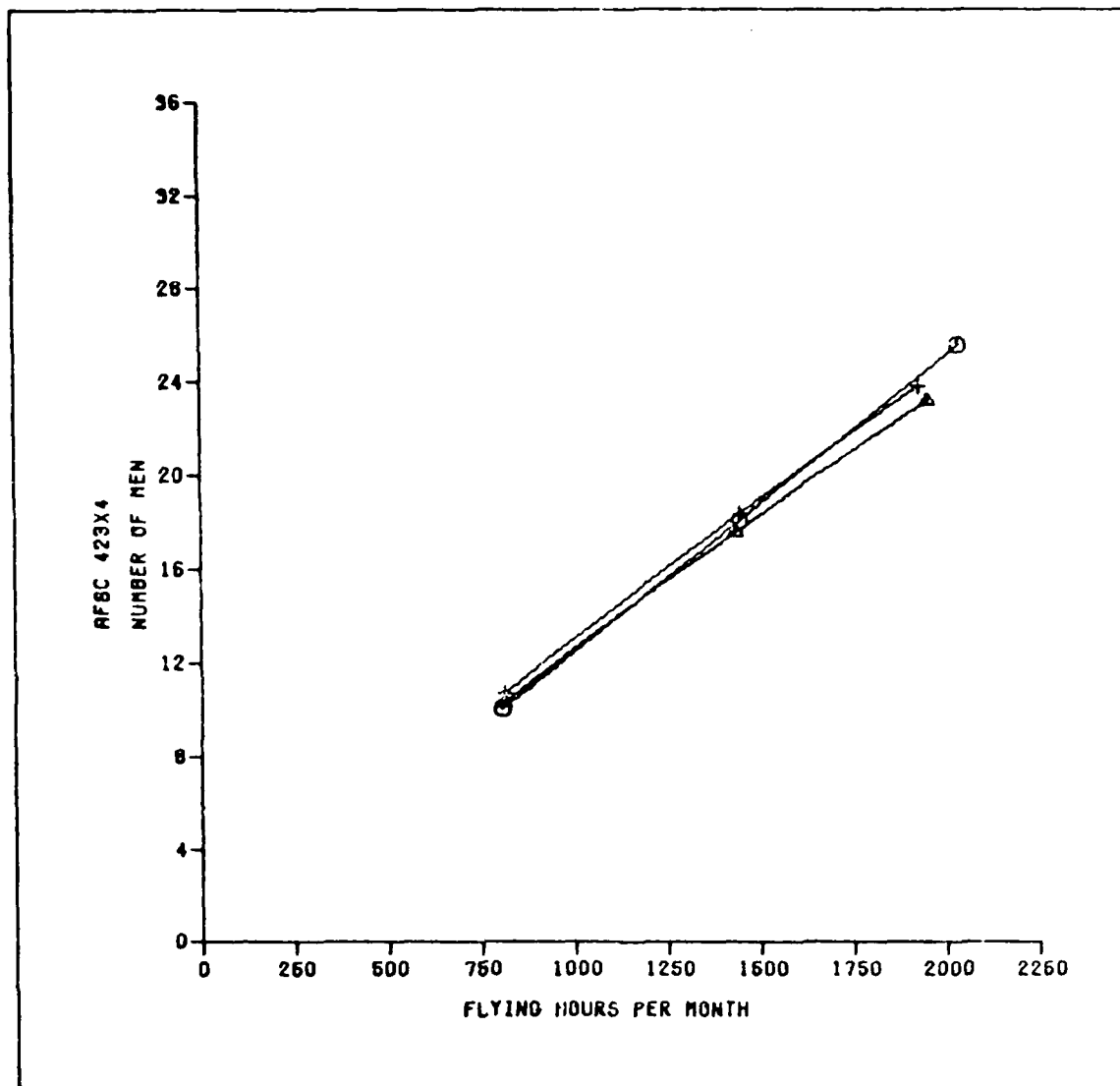
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
○ - Unconstrained Parts/ Unconstrained ATS	7 (5-9)	16 (14-17)	18 (16-19)
Δ - Constrained Parts/ Unconstrained ATS	5 (5-6)	10 (10-11)	16 (15-18)
+ - Constrained Parts/ Constrained ATS	7 (6-8)	9 (8-9)	12 (11-13)

Figure 21. AFSC 423X3 (Fuel Systems) Direct Manning



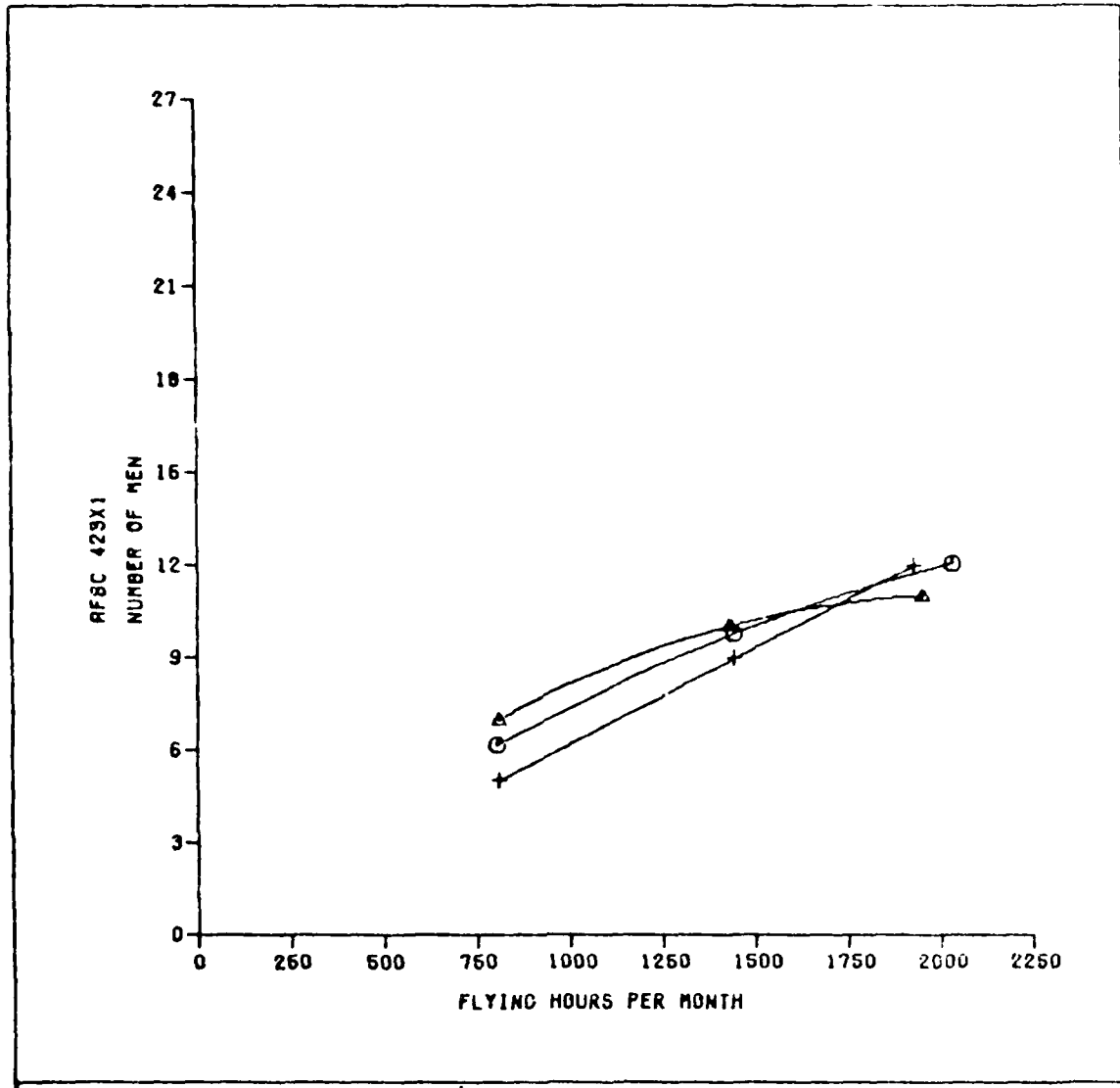
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
O - Unconstrained Parts/ Unconstrained ATS	5 (5-5)	8 (7-8)	10 (9-10)
Δ - Constrained Parts/ Unconstrained ATS	4 (4-5)	7 (6-7)	9 (9-10)
+ - Constrained Parts/ Constrained ATS	4 (4-4)	7 (7-7)	10 (9-10)

Figure 22. AFSC 423X0 (Electrical Systems) Direct Manning



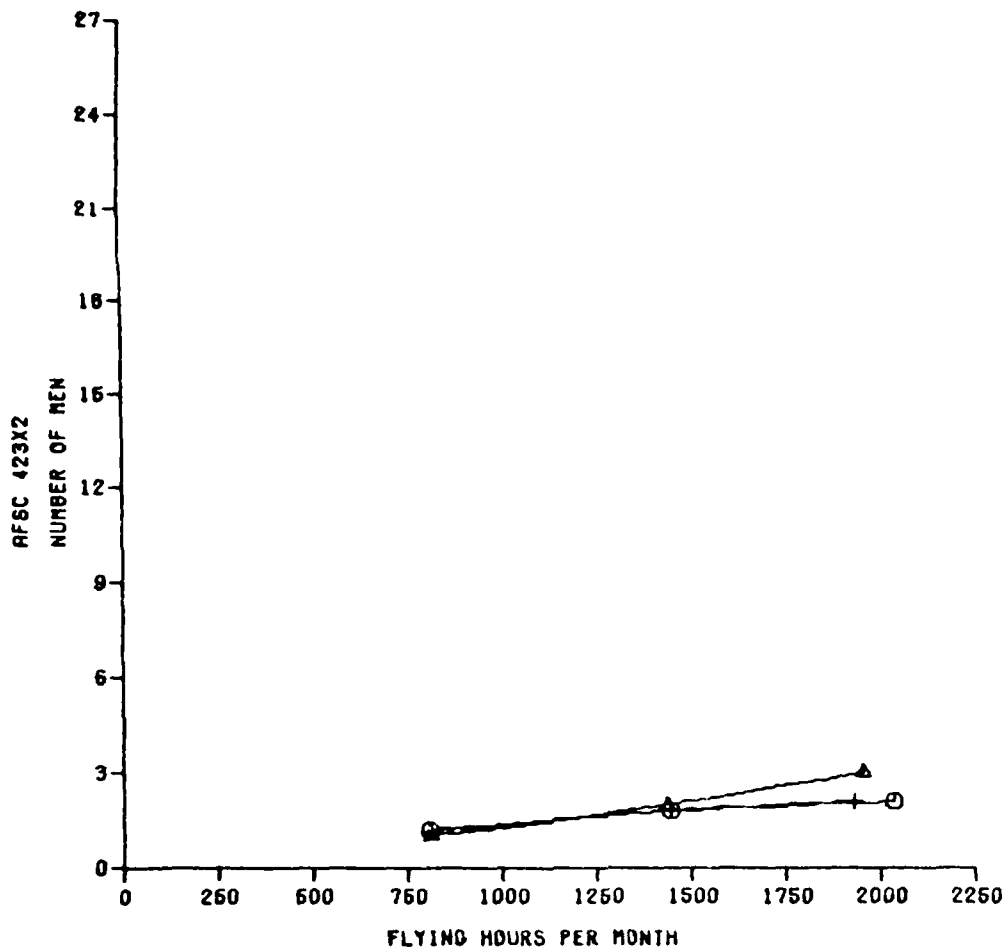
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
0 - Unconstrained Parts/ Unconstrained ATS	11 (10-11)	17 (16-19)	26 (25-27)
Δ - Constrained Parts/ Unconstrained ATS	10 (9-11)	18 (17-19)	23 (22-24)
+ - Constrained Parts/ Constrained ATS	11 (10-11)	18 (18-19)	24 (22-25)

Figure 23. AFSC 423X4 (Pneudraulics) Direct Manning



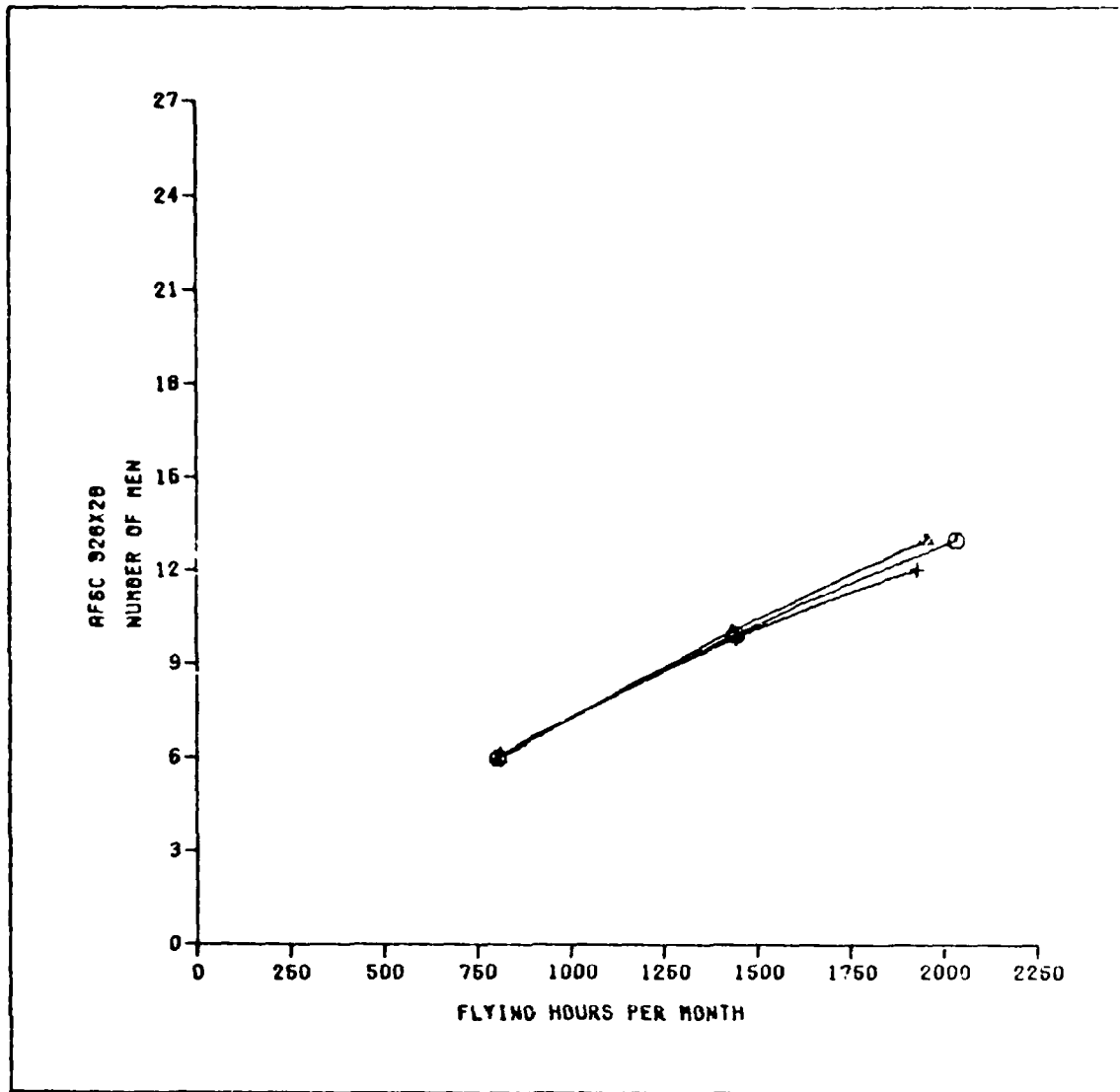
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
O - Unconstrained Parts/ Unconstrained ATS	6 (5-6)	10 (9-10)	12 (12-13)
Δ - Constrained Parts/ Unconstrained ATS	7 (7-7)	10 (9-11)	11 (10-11)
+ - Constrained Parts/ Constrained ATS	5 (5-5)	9 (9-9)	12 (11-13)

Figure 24. AFSC 423X1 (Environmental Systems) Direct Manning



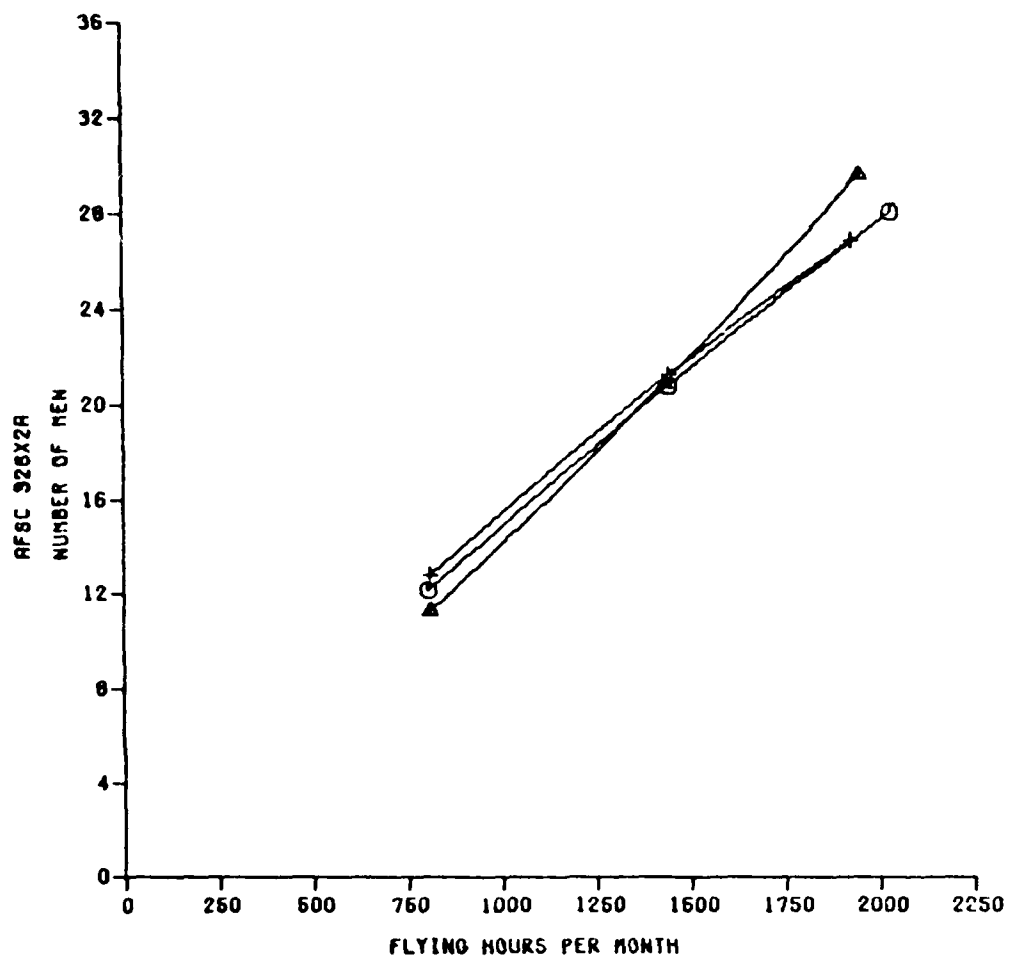
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
0 - Unconstrained Parts/ Unconstrained ATS	1 (1-2)	2 (2-2)	2 (2-3)
Δ - Constrained Parts/ Unconstrained ATS	1 (1-1)	2 (2-2)	3 (2-3)
+ - Constrained Parts/ Constrained ATS	1 (1-2)	2 (1-2)	2 (2-3)

Figure 25. AFSC 423X2 (Egress Systems) Direct Manning



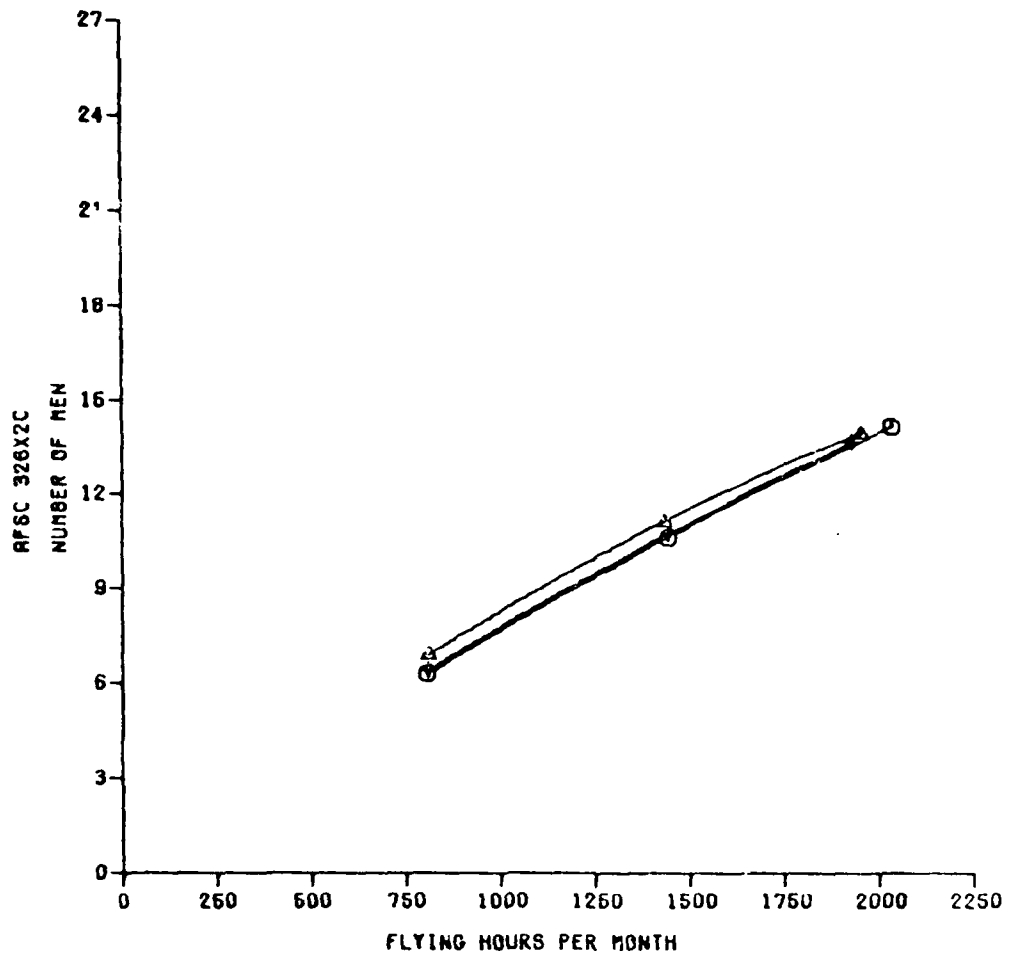
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
O - Unconstrained Parts/ Unconstrained ATS	6 (5-6)	10 (9-10)	13 (13-13)
Δ - Constrained Parts/ Unconstrained ATS	6 (5-6)	10 (9-10)	13 (13-14)
+ - Constrained Parts/ Constrained ATS	6 (5-6)	10 (10-10)	12 (12-12)

Figure 26. AFSC 326X2B (Automatic Flight Control/Instrument) Direct Manning



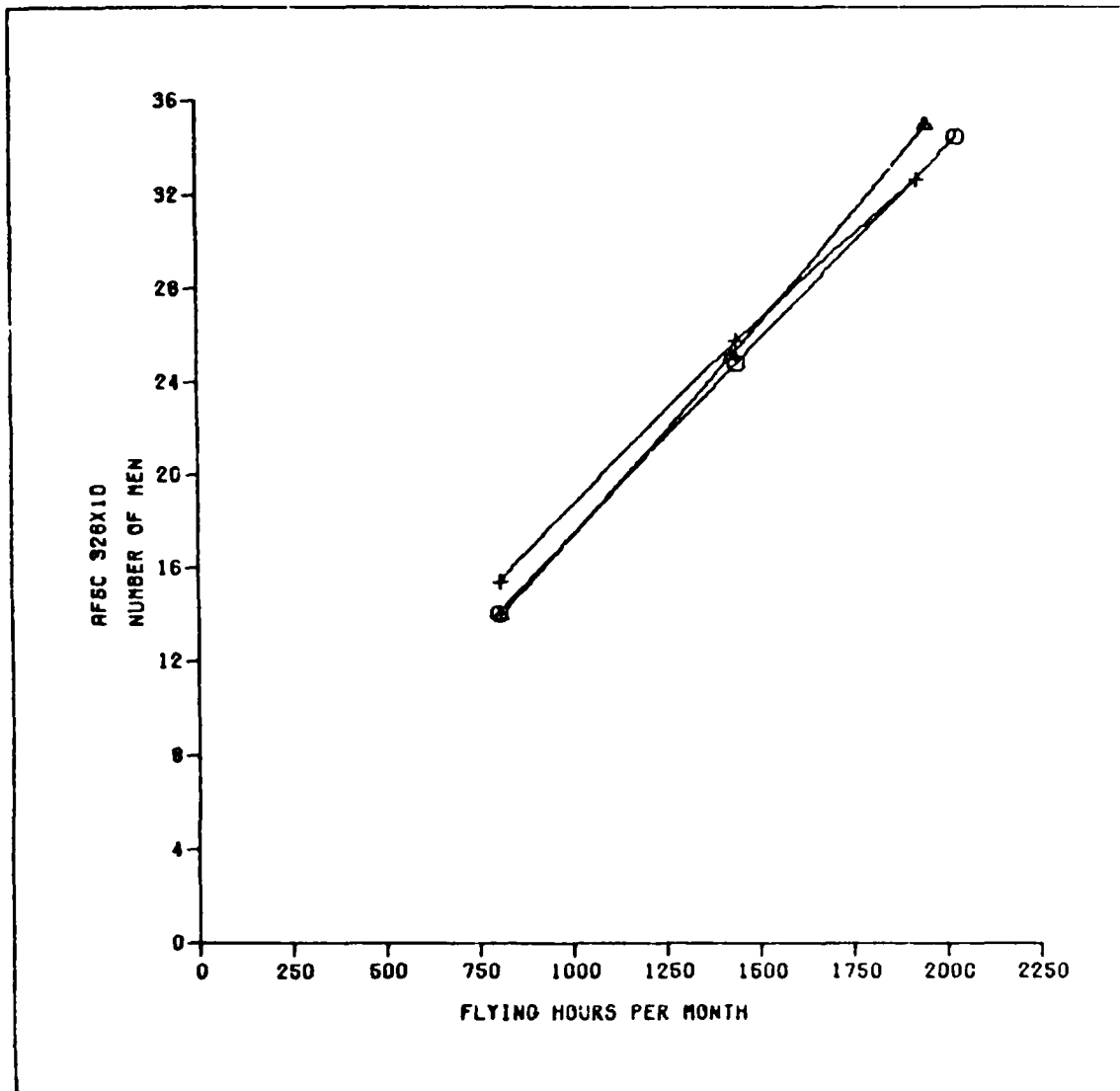
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
○ - Unconstrained Parts/ Unconstrained ATS	12 (11-12)	21 (21-22)	28 (28-29)
△ - Constrained Parts/ Unconstrained ATS	12 (12-12)	20 (19-21)	30 (23-31)
+ - Constrained Parts/ Constrained ATS	13 (12-13)	21 (20-22)	27 (25-30)

Figure 27. AFSC 326X2A (Inertial Navigation System/Weapon Control)
Direct Manning



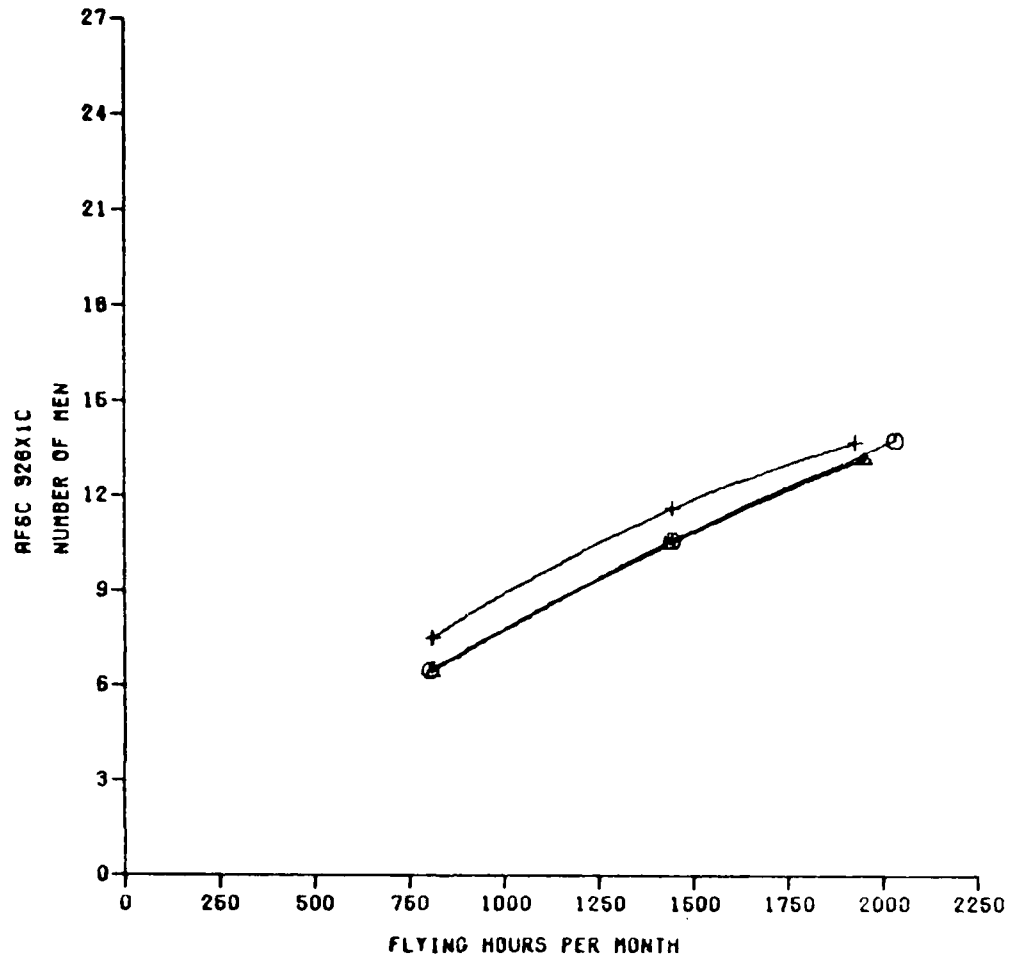
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
0 - Unconstrained Parts/ Unconstrained ATS	6 (6-7)	11 (10-11)	14 (13-14)
Δ - Constrained Parts/ Unconstrained ATS	7 (6-7)	11 (10-11)	14 (14-15)
+ - Constrained Parts/ Constrained ATS	7 (6-7)	10 (10-11)	14 (13-15)

Figure 28. AFSC 326X2C (Communications/Navigations/Electronic Counter Measures) Direct Manning



Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
○ - Unconstrained Parts/ Unconstrained ATS	13 (13-14)	26 (24-27)	34 (33-36)
△ - Constrained Parts/ Unconstrained ATS	14 (13-15)	25 (23-27)	35 (34-37)
+ - Constrained Parts/ Constrained ATS	16 (15-16)	25 (25-26)	33 (31-34)

Figure 29. AFSC 326X1D (Automatic Test Station) Direct Manning



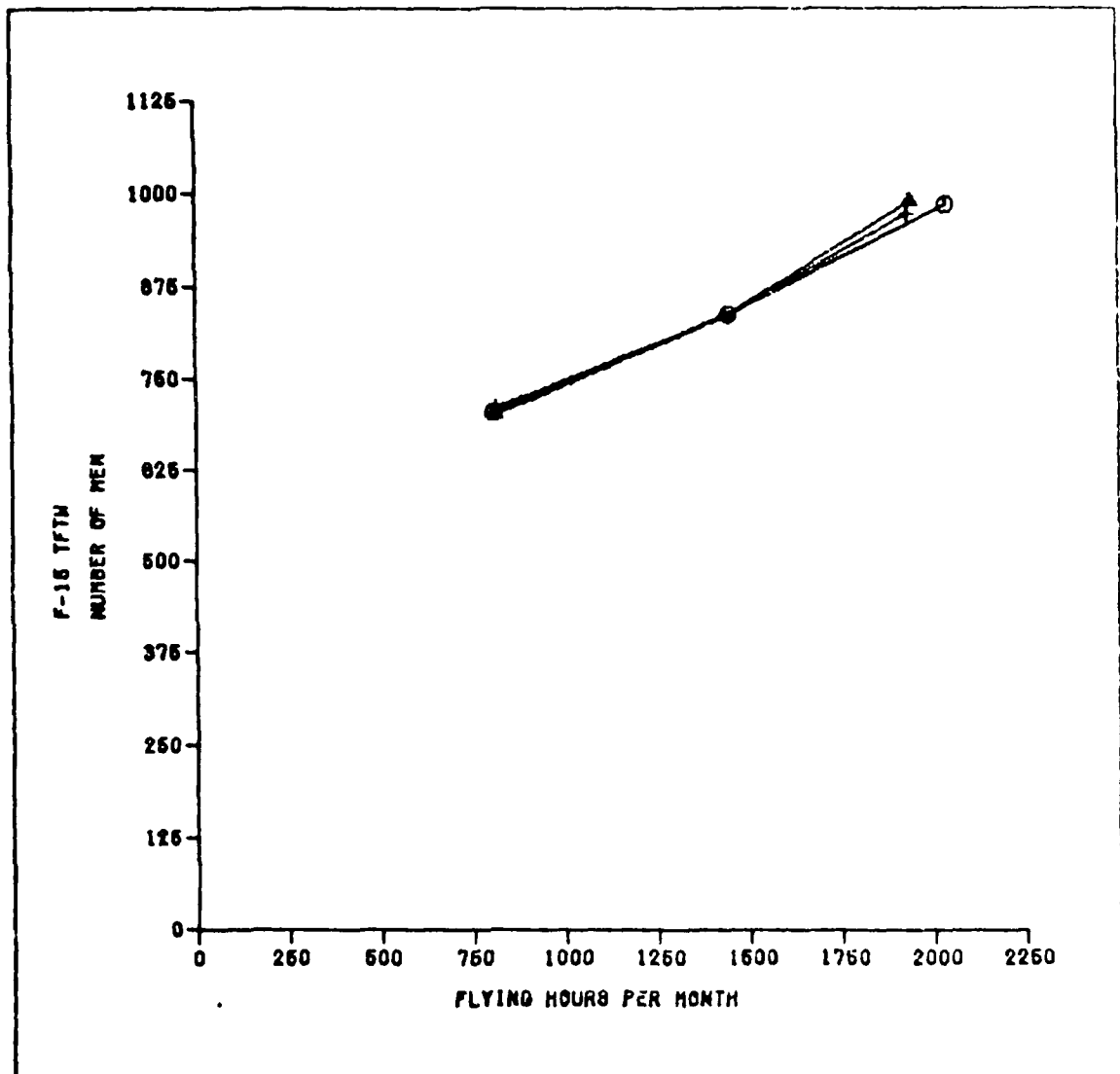
Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
0 - Unconstrained Parts/ Unconstrained ATS	7 (7-8)	10 (9-11)	14 (13-14)
Δ - Constrained Parts/ Unconstrained ATS	6 (6-7)	11 (9-12)	13 (12-13)
+ - Constrained Parts/ Constrained ATS	8 (7-8)	11 (10-11)	14 (12-16)

Figure 30. AFSC 326X1C (Manual Test Station) Direct Manning

Manning Document. The authors used the Moody Manpower Program to develop a complete basic manning document for the maintenance organization depicted in Figure 6 of Chapter III. Additionally, the program was used to determine the total maintenance manning requirements for each constraint type and scheduled sortie rate. The Moody Manpower Program considers each AFSC's minimum crew manning during the development of total manning requirements. If direct manning estimates are below the respective minimum crew manning, the program uses the minimum crew manning to determine total manning requirements. Since the LCOM networks define minimum crew requirements for each maintenance task, AFSC minimum crew manning remains constant for each constraint type and scheduled sortie rate. Table IX depicts the minimum crew manning for each AFSC. Figure 31 indicates the sensitivity of the F-15 TFTW's total manning requirements to constraint type and scheduled sortie rate. Figure 32 illustrates the basic manning document for unconstrained parts/ATS at a .74 scheduled sortie rate.

Table IX
AFSC Minimum Crew Manning

AFSC	Work Description	Minimum Crew Manning
431X1	Flight Line Crew Chief	31
431X1	Phase Inspection	24
531X3	Structural Repair	10
426X2	Jet Engine	48
423X3	Fuel Systems	10
423X0	Electrical Systems	10
423X4	Pneudraulics	10
423X1	Environmental Systems	10
423X2	Egress Systems	10
326X2B	Automatic Flight Control/Instruments	10
326X2A	Inertial Navigation System/Weapon Control	10
326X2C	Communications/Navigations/Electronic Counter Measures	10
326X1D	Automatic Test Station	10
326X1C	Manual Test Station	10



Constraint Type	Scheduled Sortie Rate		
	.43	.74	1.0
0 - Unconstrained Parts/ Unconstrained ATS	705	837	984
Δ - Constrained Parts/ Unconstrained ATS	703	834	990
+ - Constrained Parts/ Constrained ATS	710	836	973

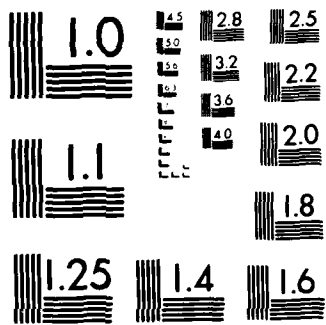
Figure 31. F-15 TFW Total Manning Requirements

F15 TFW MANNING DOCUMENT

CHIEF OF MAINTENANCE

F/C	DESC.	OSC	AFSC	GRADE	RQMT
210000	CHIEF OF MAINT	JN	4095	COL	1
210000	CHIEF OF MAINT	JN	4016	LTC	1
210000	CHIEF OF MAINT	JN	43191	CMS	1
210000	CHIEF OF MAINT	JN	43171C	MSG	1
210000	CHIEF OF MAINT	JN	70450	CIV	1
					6
210000	CHIEF OF MAINT	JND	70270	MSG	1
210000	CHIEF OF MAINT	JND	70250	SSG	2
210000	CHIEF OF MAINT	JND	70250	SGT	3
210000	CHIEF OF MAINT	JND	70230	A1C	1
					7
210000	CHIEF OF MAINT ENG MGT	JNB	75193	SMS	1
210000	CHIEF OF MAINT ENG MGT	JNB	75172	TSG	1
210000	CHIEF OF MAINT ENG MGT	JNB	75132	SSG	1
					3
210000	CHIEF OF MAINT PROD ANAL	JNA	39170A	MSG	1
210000	CHIEF OF MAINT PROD ANAL	JNA	39150A	SSG	1
210000	CHIEF OF MAINT PROD ANAL	JNA	39150A	SGT	1
					3
210000	QUALITY CONT ADM	JND	70250	CIV	2
					2
210000	PROGRAMS/MOBILITY	JNE	66170	MSG	2
					2
211000	QUALITY CONT FCF	JNHA	F4024	CPT	2
211000	QUALITY CONT	JNHA	43171C	TSG	1
					3
211000	QUALITY CONT I O DEST	JNHC	43151C	SSG	2
					2
211000	QUALITY CONT	JNH	4016	LTC	1
211000	QUALITY CONT	JNH	43191	SMS	1
					2

Figure 32. F-15 TFW Manning Document



MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

211000	QUALITY CONT	JNHB	42375	MSG	1
211000	QUALITY CONT	JNHR	42375	TSG	2
211000	QUALITY CONT	JNHR	46272	MSG	1
211000	QUALITY CONT	JNHB	46272	TSG	4
211000	QUALITY CONT	JNHB	43171C	MSG	2
211000	QUALITY CONT	JNHR	43171C	TSG	4
211000	QUALITY CONT	JNHR	43151C	SSG	1
211700	QUALITY CONT	JNHB	46270	MSG	1
211000	QUALITY CONT	JNHR	46270	TSG	1
211000	QUALITY CONT	JNHB	46170	MSG	1
211000	QUALITY CONT	JNHR	46170	TSG	1
211000	QUALITY CONT	JNHB	32672A	MSG	1
211000	QUALITY CONT	JNHR	32672A	TSG	1
211000	QUALITY CONT	JNHB	326729	TSG	1
211000	QUALITY CONT	JNHR	32672C	MSG	1
211000	QUALITY CONT	JNHB	32672C	TSG	2

25

212000	MAINT CONT	JNI	4016	LTC	1
212000	MAINT CONT	JNI	4024	CPT	1
212000	MAINT CONT	JNI	39290	CMS	1
212000	MAINT CONT	JNI	70250	SSG	1

4

212000	MAINT CONT JOB CNTL	JNIA	43191	SMS	1
212000	MAINT CONT JOB CNTL	JNIA	43171C	MSG	2
212000	MAINT CONT JOB CNTL	JNIA	43171C	TSG	6
212000	MAINT CONT JOB CNTL	JNIA	43151C	SSG	7
212000	MAINT CONT JOB CNTL	JNIA	43151C	SGT	8
212000	MAINT CONT JOB CNTL	JNIA	42355	SSG	2
212000	MAINT CONT JOB CNTL	JNIA	42355	SGT	2
212000	MAINT CONT JOB CNTL	JNIA	32672A	MSG	1
212000	MAINT CONT JOB CNTL	JNIA	32652A	SSG	1
212000	MAINT CONT JOB CNTL	JNIA	32652A	SGT	1
212000	MAINT CONT JOB CNTL	JNIA	46270	MSG	1
212000	MAINT CONT JOB CNTL	JNIA	46270	TSG	1
212000	MAINT CONT JOB CNTL	JNIA	46250	SSG	1
212000	MAINT CONT JOB CNTL	JNIA	46250	SGT	1

35

212000	MAINT CONT PLANS/SCH/000	JNIR	32672A	MSG	1
212000	MAINT CONT PLANS/SCH/000	JNIR	32672B	TSG	1
212000	MAINT CONT PLANS/SCH/000	JNIR	43171C	TSG	3
212000	MAINT CONT PLANS/SCH/000	JNIB	39230	SSG	6
212000	MAINT CONT PLANS/SCH/000	JNIR	39230	SGT	2
212000	MAINT CONT PLANS/SCH/000	JNIB	39270	MSG	2
212000	MAINT CONT PLANS/SCH/000	JNIR	39270	TSG	4
212000	MAINT CONT PLANS/SCH/000	JNIR	39290	SMS	1
212000	MAINT CONT PLANS/SCH/000	JNIR	70250	SGT	1
212000	MAINT CONT PLANS/SCH/000	JNIB	43171C	TSG	2
212000	MAINT CONT PLANS/SCH/000	JNIR	43151C	SSG	1
212000	MAINT CONT PLANS/SCH/000	JNIR	39270	MSG	1
212000	MAINT CONT PLANS/SCH/000	JNIR	39230	SSG	1

26

212000	MAINT CONT MAT CONT	JNIC	6424A	CPT	1
212000	MAINT CONT MAT CONT	JNIC	64570	MSG	1
212000	MAINT CONT MAT CONT	JNIC	70250	SGT	1

3

212000	MAINT CONT MAT CONT	JNICA	64570	TSG	4
212000	MAINT CONT MAT CONT	JNICA	64550	SSG	4
212000	MAINT CONT MAT CONT	JNICA	64550	SGT	5
212000	MAINT CONT MAT CONT	JNICA	64530	AIC	3

16

212000	MAINT CONT MAT CONT	JNICB	39270	MSG	3
212000	MAINT CONT MAT CONT	JNICB	39230	SSG	6
212000	MAINT CONT MAT CONT	JNICB	39230	SGT	6
212000	MAINT CONT MAT CONT	JNICB	39270	TSG	3

18

Figure 32. F-15 TFTW Manning Document (continued)

ORGANIZA. MAINTENANCE SQUADRON

F/C	DESC.	OSC	AFSC	GRADE	RQMT.
220000	ORGANIZ. MAINT SQ	AA	4016	MAJ	1
220000	ORGANIZ. MAINT SQ	AA	44016	LTG	1
220000	ORGANIZ. MAINT SQ	AA	70250	SSG	1
220000	ORGANIZ. MAINT SQ	AA	70230	A1C	1
220000	ORGANIZ. MAINT SQ	AA	43191	CMS	1
					5
220000	UNIT ADMIN	AU	7034	CPT	1
220000	UNIT ADMIN	AU	70250	SSG	1
220000	UNIT ADMIN	AU	70230	A1C	2
220000	UNIT ADMIN	AU	10090	MSG	1
					5
221000	FLIGHT LINE SUPV	MLM	4024	CPT	1
221000	FLIGHT LINE SUPV	MLM	70230	A1C	1
221000	FLIGHT LINE SUPV	MLM	4024	LT	1
221000	FLIGHT LINE SJPV	MLM	43191	SMS	1
221000	FLIGHT LINE CMF	MLM	43171C	MSG	1
221000	FLIGHT LINE CMF	MLM	43171C	MSG	1
221000	FLIGHT LINE CMF	MLM	43171C	MSG	4
221000	FLIGHT LINE CMF	MLM	43171C	MSG	4
221000	FLIGHT LINE CMF	MLM	43171C	TSG	20
221000	FLIGHT LINE EXPEDITOR	MLM	43171C	TSG	9
221000	FLIGHT LINE CMF	MLM	43151C	SSG	4
221000	FLIGHT LINE INSPECTOR	MLM	43191	SMS	1
221000	FLIGHT LINE GO SUPT EDP	MLM	43171C	MSG	1
221000	FLIGHT LINE BENCH STOCK	MLM	43151C	SSG	1
221000	FLIGHT LINE BENCH STOCK	MLM	43151C	SGT	2
221000	FLIGHT LINE 700 EQUIP	MLM	43171C	TSG	1
221000	FLIGHT LINE 700 EQUIP	MLM	43151C	SSG	1
221000	FLIGHT LINE 700 EQUIP	MLM	43151C	SGT	1
221000	FLIGHT LINE 700 EQUIP	MLM	43131C	A1C	1
221000	FLIGHT LINE	MLM	43191	SMS	1
221000	FLIGHT LINE	MLM	43191	MSG	1
221000	FLIGHT LINE	MLM	43171C	MSG	4
221000	FLIGHT LINE	MLM	43171C	TSG	1
221000	FLIGHT LINE	MLM	43171C	SSG	5
221000	FLIGHT LINE	MLM	43151C	SSG	19
221000	FLIGHT LINE	MLM	43151C	SGT	41
221000	FLIGHT LINE	MLM	43151C	A1C	10
221000	FLIGHT LINE	MLM	43131C	A1C	41
					176
222000	PHASE	MLI	431	71CMSG	1
222000	PHASE	MLI	431	71GTSG	1
222000	PHASE	MLI	431	71CSSG	1
222000	PHASE	MLI	431	51C5SG	4
222000	PHASE	MLI	431	51C5GT	8
222000	PHASE	MLI	431	51CA1C	2
222000	PHASE	MLI	431	31CA1C	7
					24

Figure 32. F-15 TFTW Manning Document (continued)

FIELD MAINTENANCE SQUADRON

F/C	DESC.	OSC	AFSC	GRADE	RQMT
230000	FIELD MAINT SQ	AA	4015	MAJ	1
230000	FIELD MAINT SQ	AA	44015	LTC	1
230000	FIELD MAINT SQ	AA	70250	SSG	1
230000	FIELD MAINT SQ	AA	42692	CMS	1
230000	FIELD MAINT SQ	AA	70230	A1C	1
					5
230000	UNIT ADMIN	AU	7034	CPT	1
230000	UNIT ADMIN	AU	70250	SSG	1
230000	UNIT ADMIN	AU	70230	A1C	1
230000	UNIT ADMIN	AU	10090	MSG	1
					4
231000	FABRICATION	MKM	53195	SMS	1
					1
231100	MACHINE SHOP	MKMA	53150	SSG	2
231100	MACHINE SHOP	MKMA	53150	SGT	2
231100	MACHINE SHOP	MKMA	53150	A1C	1
231100	MACHINE SHOP	MKMA	53130	A1C	1
					6
231200	METAL PROCESSING	MKMB	53151	SSG	2
231200	METAL PROCESSING	MKMB	53151	SGT	2
231200	METAL PROCESSING	MKMB	53131	A1C	2
					6
231300	STRUT REPAIR	MKMC	53173	SSG	1
231300	STRUT REPAIR	MKMC	53153	SSG	2
231300	STRUT REPAIR	MKMC	53153	SGT	5
231300	STRUT REPAIR	MKMC	53153	A1C	1
231300	STRUT REPAIR	MKMC	53133	A1C	5
					14
231400	CORROSION CONTROL	MKMF	53154	SSG	1
231400	CORROSION CONTROL	MKMF	53154	SGT	2
231400	CORROSION CONTROL	MKMF	53154	A1C	3
231400	CORROSION CONTROL	MKMF	53134	A1C	2
					8
231500	SURVIVAL EQUIPMENT	MKME	50271	TSG	1
231500	SURVIVAL EQUIPMENT	MKME	50251	SSG	2
231500	SURVIVAL EQUIPMENT	MKME	50251	SGT	2
231500	SURVIVAL EQUIPMENT	MKME	50231	A1C	2
231500	SURVIVAL EQUIPMENT	MKME	50250	SSG	1
231500	SURVIVAL EQUIPMENT	MKME	50250	SGT	1
231500	SURVIVAL EQUIPMENT	MKME	50230	A1C	1
					10
231700	NON-DESTRUCTIVE INSPECTION	MKMG	53175	CIV	1
231700	NON-DESTRUCTIVE INSPECTION	MKMG	53175	TSG	1
231700	NON-DESTRUCTIVE INSPECTION	MKMG	53155	SSG	2
231700	NON-DESTRUCTIVE INSPECTION	MKMG	53155	SGT	2
231700	NON-DESTRUCTIVE INSPECTION	MKMG	53135	A1C	2
					8
232000	PROPULSION	MKI	4024	LT	1
232000	PROPULSION	MKI	45292	CMS	1
232000	BENCH STOCK/TOOL RM	MKI	42672	TSG	1
232000	BENCH STOCK/TOOL RM	MKI	42652	SSG	1
232000	BENCH STOCK/TOOL RM	MKI	42652	SGT	1
232000	BENCH STOCK/TOOL RM	MKI	42632	A1C	1
					6

Figure 32. F-15 TFTW Manning Document (continued)

232330	JET ENGINE	MKIC	42672	MSG	2
232300	JET ENGINE	MKIC	42672	TSG	2
232300	JET ENGINE	MKIC	42672	SSG	4
232330	JET ENGINE	MKIC	42652	SSG	16
232300	JET ENGINE	MKIC	42652	SGT	34
232300	JET ENGINE	MKIC	42652	AIC	8
232300	JET ENGINE	MKIC	42632	AIC	35
					101
233030	AEROSPACE SYSTEMS	MKJ	4024	LT	1
233000	AEROSPACE SYSTEMS	MKJ	70250	SGT	1
233000	AEROSPACE SYSTEMS	MKJ	42396	SMS	1
					3
233100	REPAIR AND RECLAMATION	MKJA	431710	MSG	2
233100	REPAIR AND RECLAMATION	MKJA	431710	TSG	3
233100	REPAIR AND RECLAMATION	MKJA	431510	SSG	6
233100	REPAIR AND RECLAMATION	MKJA	431510	SGT	6
233100	REPAIR AND RECLAMATION	MKJA	431310	AIC	5
					22
233200	FUEL SYSTEMS	MKJB	42373	SSG	1
233200	FUEL SYSTEMS	MKJB	42353	SSG	2
233200	FUEL SYSTEMS	MKJB	42353	SGT	5
233200	FUEL SYSTEMS	MKJB	42353	AIC	1
233200	FUEL SYSTEMS	MKJB	42333	AIC	5
					14
233300	ELECTRICAL SYSTEMS	MKJC	42370	SSG	1
233300	ELECTRICAL SYSTEMS	MKJC	42350	SSG	2
233300	ELECTRICAL SYSTEMS	MKJC	42350	SGT	1
233300	ELECTRICAL SYSTEMS	MKJC	42350	AIC	1
233300	ELECTRICAL SYSTEMS	MKJC	42330	AIC	3
					10
233400	PNEUMRAULICS	MKJD	42374	SSG	1
233401	PNEUMRAULICS	MKJD	42354	SSG	3
233400	PNEUMRAULICS	MKJD	42354	SGT	6
233401	PNEUMRAULICS	MKJD	42354	AIC	1
233400	PNEUMRAULICS	MKJD	42334	AIC	7
					18
233600	ENVIRONMENTAL SYSTEMS	MKJF	42351	SSG	2
233600	ENVIRONMENTAL SYSTEMS	MKJF	42361	SGT	1
233600	ENVIRONMENTAL SYSTEMS	MKJF	42351	AIC	1
233600	ENVIRONMENTAL SYSTEMS	MKJF	42331	AIC	4
					10
233900	EGRESS SYSTEMS	MKJI	42352	SSG	2
233900	EGRESS SYSTEMS	MKJI	42352	SGT	3
233900	EGRESS SYSTEMS	MKJI	42352	AIC	1
233900	EGRESS SYSTEMS	MKJI	42332	AIC	4
					10
234000	AEROSPACE GROUND EQUIPMENT	MKK	42375	MSG	2
234000	AEROSPACE GROUND EQUIPMENT	MKK	42375	TSG	2
234000	AEROSPACE GROUND EQUIPMENT	MKK	42375	SSG	4
234000	AEROSPACE GROUND EQUIPMENT	MKK	42355	SSG	15
234000	AEROSPACE GROUND EQUIPMENT	MKK	42355	SGT	33
234000	AEROSPACE GROUND EQUIPMENT	MKK	42355	AIC	8
234000	AEROSPACE GROUND EQUIPMENT	MKK	42335	AIC	34
					98

Figure 32. F-15 TFW Manning Document (continued)

AVIONICS COMMAND SQUADRON

F/C	DESC.	OSC	AFSC	GRADE	RQMT
240000	AVIONICS SQ	AA	44096	LTC	1
240000	AVIONICS SQ	AA	4016	MAJ	1
240000	AVIONICS SQ	AA	32090	CMS	1
240000	AVIONICS SQ	AA	70250	SSG	1
240000	AVIONICS SQ	AA	70230	A1C	1
240000	AVIONICS SQ	AA	4024	CPT	1
					6
240000	UNIT ADMIN	AU	7034	CPT	1
240000	UNIT ADMIN	AU	10090	MSG	1
240000	UNIT ADMIN	AU	70250	SSG	1
240000	UNIT ADMIN	AU	70230	A1C	1
					4
243000	ELECTRONICS	HJO	4024	CPT	1
243000	ELECTRONICS	HJO	32672C	MSG	1
					2
243300	AUT FLT CON & INST SUPV	HJQ	32692	SMS	1
243300	AUT FLT CON & INST	HJQ	32652B	SSG	2
243300	AUT FLT CON & INST	HJQ	32652B	SGT	1
243300	AUT FLT CON & INST	HJQ	32652B	A1C	1
243300	AUT FLT CON & INST	HJQ	32632M	A1C	4
					11
243600	WEAP CON & INER NV SJPV	HJOA	32692	SMS	1
243600	WEAP CON & INER NV	HJOA	32672A	MSG	1
243600	WEAP CON & INER NV	HJOA	32672A	TSG	1
243600	WEAP CON & INER NV	HJOA	32672A	SSG	1
243600	WEAP CON & INER NV	HJOA	32652A	SSG	3
243600	WEAP CON & INER NV	HJOA	32652A	SGT	7
243600	WEAP CON & INER NV	HJOA	32652A	A1C	1
243600	WEAP CON & INER NV	HJOA	32632A	A1C	6
					21
243700	COMM NV & PEN AIDS	HJOC	32652C	SSG	1
243700	COMM NV & PEN AIDS	HJOC	32652C	SGT	3
243700	COMM NV & PEN AIDS	HJOC	32652C	A1C	1
243700	COMM NV & PEN AIDS	HJOC	32632C	A1C	5
					10
246000	AVN SHOP MAINT	HJP	4024	CPT	1
246000	AVN SHOP MAINT	HJP	32692	MSG	1
					2

Figure 32. F-15 TFTW Manning Document (continued)

246100	AVN AGE	HJPA	32670R	MSG	1
246100	AVN AGE	HJPA	32670B	TSG	1
246100	AVN AGE	HJPA	32679H	SSG	1
246100	AVN AGE	HJPA	32650R	SSG	4
246100	AVN AGE	HJPA	32650R	SGT	9
246100	AVN AGE	HJPA	32650R	A1C	2
246100	AVN AGE	HJPA	32630B	A1C	8

26

246200	AUTO TST STN SUPV	HJPB	32692	SMS	1
246200	AUTO TST STN	HJPR	326710	MSG	1
246200	AUTO TST STN	HJPB	326710	TSG	1
246200	AUTO TST STN	HJPR	326710	SSG	1
246200	AUTO TST STN	HJPR	326510	SSG	4
246200	AUTO TST STN	HJPA	326510	SGT	6
246200	AUTO TST STN	HJPR	326510	A1C	2
246200	AUTO TST STN	HJPB	326310	A1C	7

26

246300	MAN TST STN	HJPC	32651C	SSG	1
246300	MAN TST STN	HJPC	32651C	SGT	3
246300	MAN TST STN	HJPC	32651C	A1C	1
246300	MAN TST STN	HJPC	32631C	A1C	2

10

SUMMARY OF WHOLE MING

COL	1
LTC	6
MAJ	3
CPT	11
LT	3
TOTAL	24

GIV	4

CMS	6
SMS	12
MSG	5
MSG	46
TSG	89
SSG	23
SSG	146
SGT	227
A1C	47
A1C	20A
TOTAL	599

Figure 32. F-15 TFTW Manning Document (continued)

Summary

This chapter contains a discussion of the analysis and results of the F-15 TFTW LCOM simulation. The analysis section contains a graphical and statistical interpretation of the simulation's steady state conditions and autocorrelation coefficients. The results consist of direct manning estimates for each AFSC listed in Table I, their respective 95 percent confidence intervals, the sensitivity of these estimates to variations in parts and ATS, and a USAF basic manning document for the F-15 TFTW. The Conclusion and Recommendation Chapter summarizes the results and makes recommendations concerning the thesis findings.

VI. CONCLUSIONS AND RECOMMENDATIONS

As stated in Chapter I, this thesis had three major objectives: first, use LCOM to estimate the maintenance manpower requirements for a 72 U.E. F-15 TFTW; second, construct statistical confidence intervals for these manpower estimates; third, investigate the sensitivity of maintenance manpower requirements to variations in the availability of aircraft spare parts and support equipment. The authors accomplished these objectives and presented their analysis and results in Chapter V. Figures 17 through 30 illustrate each AFSC's direct manning requirements, confidence intervals, and sensitivity to constraint type and scheduled sortie rate. Figure 31 depicts the total manning requirements for a 72 U.E. F-15 TFTW and the sensitivity of these requirements to constraint type and scheduled sortie rate. Finally, Figure 32 contains a basic manning document for a 72 U.E. F-15 TFTW based on unconstrained parts/ATS and a .74 scheduled sortie rate.

Conclusions

This thesis has allowed the authors to develop several noteworthy conclusions concerning LCOM manpower estimation, in general, and the peacetime model used in this study. They feel that these developments will enable greater statistical accuracy in LCOM manpower estimates and, therefore, further enhance LCOM's position in the manpower community.

Steady State. In this study, the authors developed a more efficient procedure for determining steady state conditions in an LCOM computer simulation. The procedure involved graphical and statistical analysis of daily and weekly AFSC manhour totals as described in Chapters IV and V.

Using this technique, the authors found that the F-15 TFW model used in this thesis exhibited steady state conditions throughout the 98 day simulation period. In those isolated instances found in Table IV of Chapter V where individual AFSC's indicated transient behavior, this behavior disappeared after the first or second week into the simulation period. As a result, output data for the entire simulation period was used to estimate maintenance manpower requirements. When contrasted with the more common procedure of discarding the first 30 days in an LCOM simulation (Ref 33:114-115), the procedure used in this study more efficiently utilizes computer run time and allows equally accurate results. When one considers that this study required more than 25 computer runs, the overall savings in computer time is significant.

Autocorrelation. The study used autocorrelation analysis to develop 95 percent statistical confidence intervals for AFSC direct manning estimates. The procedure allowed the authors to estimate maintenance manpower requirements with a very high degree of statistical accuracy. Based on this procedure, all AFSC's except two indicated a five percent or less variability in direct manning estimates; AFSC's 426X2 (Jet Engine) and 423X3 (Fuel Systems) indicated an eight percent and nine percent, respectively, variability in direct manning estimates. Such accuracy lends confidence to the LCOM manpower estimation process.

Model Validity. In order to verify these procedures, the authors compared the LCOM direct manning estimates and corresponding statistical confidence intervals contained in Chapter V with estimates from three other LCOM simulation strategies. Table X illustrates this comparison using four sets of manning estimates based on unconstrained parts/ATS

Table X
Comparison of Simulation Strategies

AFSC	Simulation Strategy			
	1	2	3	4
431X1 (Flight Line Crew Chief)	124(120-128)	124	124	125
431X1 (Phase Inspection)	20(20-21)	21	20	20
531X3	14(13-14)	13	13	13
426X2	103(94-112)	97	107	97
423X3	16(14-17)	13	13	11
423X0	8(7-8)	7	7	8
423X4	17(16-19)	18	18	18
423X1	10(9-10)	9	10	8
423X2	2(2-2)	2	2	2
326X2B	10(9-10)	9	9	10
326X2A	21(21-22)	21	21	20
326X2C	11(10-11)	10	11	11
326X1D	26(24-27)	25	25	24
326X1C	10(9-11)	10	11	11
ASR	.697	.694	.708	.692
<p>Strategy 1 - 98 day simulation with 5 day flying schedule and 7 day maintenance schedule.</p> <p>Strategy 2 - 196 day simulation with 5 day flying schedule and 7 day maintenance schedule.</p> <p>Strategy 3 - 100 day simulation with 5 day flying schedule and 5 day maintenance schedule.</p> <p>Strategy 4 - 98 day simulation with 5 day flying schedule, 7 day maintenance schedule, and different random number seeds.</p>				

and a .74 scheduled sortie rate. In this table, Strategy 1 reflects the methodology and results contained in Chapters IV and V, respectively.

The direct manning estimates from Strategy 2 were obtained using methodology similar to Strategy 1 except the simulation period consisted of 196 days and the manning estimates were based solely on the last 98 simulated days. The reason for doubling the simulation period was to determine if the authors' assertions concerning the peacetime model's steady state behavior were, in fact, correct. Since the direct manning estimates obtained from Strategy 2 were similar to the corresponding Strategy 1 estimates, the authors concluded that longer computer run times did not significantly affect the F-15 TFTW direct manning estimates. This comparison confirmed their steady state analysis of the peacetime model.

In Strategy 3, the direct manning estimates were based on a 100 day simulation period with a concurrent five day flying and maintenance work week. Such a scenario eliminated weekends from the maintenance schedule and eliminated the opportunity to perform backlogged maintenance work on nonflying days. During the computation of direct manning under this strategy, 20.99 work days per month replaced 30.44 work days per month in Equation (3) of Chapter IV (Ref 2:Chap. 6, p. 29). Although the manning estimates from this strategy were similar to those of Strategy 1, the authors concluded that higher scheduled sortie rates, and/or more severe parts/ATS constraints would cause bottlenecks in the maintenance work flow and produce erratic day-to-day manhours used data. Such data would cause transient behavior at the beginning of the simulation period and would require longer computer run times in order to achieve steady state conditions. Since actual peacetime operations routinely schedule flying

and maintenance activity Monday through Friday and permit weekend maintenance activity to alleviate backlogged work, a scenario based on a five day flying and seven day maintenance work week is a logical simulation strategy for peacetime LCOM studies. Such a strategy reflects peacetime operations in that maintenance manpower only works on weekends when there exists a work demand and permits efficient LCOM estimation of peacetime maintenance manpower over a wide range of flying activity levels and aircraft part/ATS constraints.

The direct manning estimates from Strategy 4 were obtained using methodology identical to Strategy 1 and different random number seeds in the LCOM main program. The change in random number seeds created an independent replication of direct manning estimates; the slight differences in AFSC manning estimates between Strategies 1 and 4 are a normal occurrence in replicated data. At the same time, this small variability in AFSC direct manning between the two replications supported the internal validity of the F-15 TFTW model used in this study and reinforced the authors' confidence in the F-15 peacetime model (Ref 10:204).

It should be noted that the AFSC 423X3 (Fuel Systems) direct manning estimate in Strategies 2, 3, and 4 did not fall within the respective Strategy 1 confidence interval. Additionally, the Strategy 4 estimates for AFSC's 423X1 (Environmental Systems) and 326X2A (Inertial Navigation System/Weapon Control) did not lie within the corresponding Strategy 1 confidence intervals. The authors found these deviations interesting but not disconcerting. The authors expected isolated deviations in direct manning estimates for two reasons: first, when direct manning estimates are rounded to integer values in accordance with AFM 25-5 standards, two estimates such as 8.61 and 8.62 become eight and nine,

respectively; second, some variability in replicated data is to be expected. The comparison of simulation Strategy 1 with Strategies 2, 3, and 4 was intended to illustrate, in a general sense, that the authors methodology was valid. The authors could not completely explain the lower estimates for AFSC 423X3 in Strategies 2, 3, and 4. However, they do not feel that the deviation of these estimates from the Strategy 1 estimates was reason to question the overall validity of Strategy 1. If there had existed consistent deviations among a majority of the AFSC estimates in Strategies 2, 3, and 4 when compared to Strategy 1, the authors would have questioned the validity of their methodology. However, in general, just the opposite is true. The majority of estimates in Strategies 2, 3, and 4 agreed closely with the results of Strategy 1. Thus, the authors believe the comparison of simulation strategies supports, in a general sense, the validity of their methodology.

Manning Sensitivity. It is apparent from Figure 31 in Chapter V that F-15 TFTW total manning requirements were relatively insensitive to the constrained parts and ATS contained in Appendix E. Although total manning requirements indicated a slight sensitivity to parts and ATS constraints at the 1.0 scheduled sortie rate, these constraints had little effect on total manning requirements at the lower scheduled flying activity levels. The authors concluded that at and below a .74 scheduled sortie rate the spare part and ATS constraints contained in Appendix E were sufficient to support the daily flying activity.

At the 1.0 scheduled sortie rate, the spare parts and ATS constraints could not support the daily flying activity. As a result, the constrained parts simulation required additional manpower to repair and/or cannibalize aircraft parts while the constrained parts/ATS simulation

required less manpower because a limited number of men could perform ATS functions. As Table VIII in Chapter V indicates, these constraints also had a noticeable effect on accomplished sortie rate (ASR) and flying hours per month (FHPM). The authors concluded that more severe part and ATS constraints would have similar effects on the lower flying activity levels.

Individual AFSC manning showed no particular trend in sensitivity to constraint type. The AFSC 423X3 (Fuel Systems) in Figure 21 was the most sensitive estimate while AFSC 431X1 (Flight Line Crew Chief) displayed the least sensitivity. In most cases, the AFSC direct manning confidence intervals for the .43 and .74 scheduled sortie rates contained two or more constraint type manning estimates. The authors, therefore, concluded that, since the modeled F-15 TFTW conducts its routine flying training in the .43 and .74 scheduled sortie rate range, direct manning sensitivity to aircraft spare part and ATS constraints contained in Appendix E was statistically insignificant.

Recommendations

As a result of this study, the authors have gained an indepth understanding of the LCOM manpower estimation process. They believe this experience qualifies them to make several recommendations concerning future LCOM studies.

The authors recommend that the LCOM postprocessor program be modified to provide time series plots of AFSC manhours used. These plots would be similar to the graphs illustrated in Figures 10 through 13 in Chapter V. This capability would allow LCOM users to efficiently determine the steady state behavior of their simulation model. Tetmeyer

(Ref 31) agreed that such a capability would benefit the LCOM community; Drake (Ref 29) confirmed the feasibility of such a modification. The authors, therefore, recommend that the new version of LCOM (LCOM II) incorporate this feature.

It is recommended that LCOM peacetime studies use a five day flying and seven day maintenance work week. This type of scenario reflects actual peacetime operations and permits efficient LCOM estimation of maintenance manpower over a wide range of flying activity levels.

Finally, the authors recommend that the LCOM community use the methodology described in Chapter IV to develop statistical confidence intervals for manning estimates. This procedure adds statistical reliability to the LCOM estimates and provides decision makers with a realistic perspective of manpower requirements.

Future Work

The accomplishment of the authors stated objectives represent the conclusion of this thesis effort. However, the authors feel that future LCOM studies should investigate the interaction of peacetime manning estimates with a wartime environment. For example, the peacetime manning estimates derived from this thesis should be incorporated into the F-15 wartime model in order to measure the effect on accomplished sortie rate and manhour utilization rate. Such an effort would realistically portray the conversion of a peacetime flying organization into a wartime operation and give the decision maker direction in determining potential operational problem areas.

BIBLIOGRAPHY

1. ABD64A Data Tape. MDC Data for F-15 Aircraft. Luke Air Force Base, Arizona: September 1975 - February 1976.
2. AFM 25-5. Management Engineering Policies and Procedures. Washington: Department of the Air Force.
3. AFM 26-3. Air Force Manpower Standards. Vol I: General. Washington: Department of the Air Force.
4. AFM 66-1. Maintenance Management, Vol. I: Policy. Washington: Department of the Air Force.
5. ASD F-15/F-16 Manpower Study. HQ/USAF directed study organized by Lieutenant Colonel Donald C. Tetmeyer. Wright-Patterson Air Force Base, Ohio: Headquarters, Aeronautical Systems Division. October 1975.
6. Conway, R. W. "Some Tactical Problems in Digital Simulation," Management Science, X, 1 (1963), 47-61.
7. Drake, William F., III, Rolland R. Fisher, and John R. Younger. Logistics Composite Model Users Reference Guide. AFLC Report 70-1. Wright-Patterson Air Force Base, Ohio: Headquarters, Air Force Logistics Command, January 1970.
8. Drake, William F., III. Logistics Composite Model Users Reference Guide Update. AFLC/ADDR Report 74-1. Wright-Patterson Air Force Base, Ohio: Headquarters, Air Force Logistics Command, November 1974.
9. Drake, William F., III. AFLC/XRS. Personal Interview. Wright-Patterson Air Force Base, Ohio: 20 October 1976.
10. Emshoff, James R. and Roger L. Sisson. Design and Use of Computer Simulation Models. New York: Macmillan Publishing Co., Inc., 1970.
11. ENOI 50-8. Thesis and Dissertation. Wright-Patterson Air Force Base, Ohio: Headquarters, Air Force Institute of Technology, 30 May 1975.
12. Freund, John E. Modern Elementary Statistics. Englewood Cliffs, N. J.: Prentice-Hall, Inc., 1952.
13. Fritz, Richard and Gerald A. Yates. A Computer Simulation of Maintenance Manpower Requirements for the DC-130H. Unpublished thesis. Wright-Patterson Air Force Base, Ohio: Headquarters, Air Force Institute of Technology, June 1975.
14. Green, James C. and Paul W. Rumple. Computer Simulation of BGM-34C Maintenance Manpower Requirements. Unpublished Thesis. Wright-Patterson Air Force Base, Ohio: Air Force Institute of Technology, June 1975.

15. Gribbling, R. L., Colonel, USAF, TAC/XPMM. Letter to HQ AFIT/ENS concerning a proposed thesis topic. Langley Air Force Base, Virginia: 9 February 1976.
16. Gribbling, R. L., Colonel, USAF, TAC/XPMM. F-15 Peacetime LCOM Coordination Letter. Langley Air Force Base, Virginia: 24 February 1976.
17. Guenther, Elmer W. Jr., Colonel, USAF, TAC/LGMF. F-15 Peacetime LCOM coordination Letter, Langley Air Force Base, Virginia: 4 March 1976.
18. HQ TAC/XPM. TAC Action Plan. F-15 Peacetime LCOM Study. Langley Air Force Base, Virginia: 1 March 1976.
19. Jansen, Wayne A., ASD Computer Center. Performed LCOM Program Modifications. Wright-Patterson Air Force Base, Ohio: June 1976.
20. JarBoe, Joseph K., Lt Colonel, USAF, ASD/ENCH. Computer Service Authorization. ASD Form 145. Wright-Patterson Air Force Base, Ohio: 25 February 1976.
21. Leedom, Dennis K. and Arnold R. Thomas. A Simulation Study of the Force Mix Problem in Close Air Support Operations. ASD/XRO 73-3. Wright-Patterson Air Force Base, Ohio: Headquarters, Aeronautical Systems Division, June 1973.
22. Lowell, James R., 1st Lt, USAF, 4400MES/OLAA. Conducted an operational audit of 58th TFTW maintenance organization to verify F-15 LCOM peacetime maintenance networks. Luke Air Force Base, Arizona: March 1976.
23. Lowell, James R., 1st Lt, USAF, 4400 MES/OLAA. Personal Interview. Wright-Patterson Air Force Base, Ohio: 21 April 1976.
24. Lowell, James R., 1st Lt, USAF, and William D. Moody, 4400 MES/OLAA. Personal Interview, Wright-Patterson Air Force Base, Ohio: 27 September 1976.
25. Mendenhall, William and Richard L. Schaeffer. Mathematical Statistics with Applications. North Scituate, Massachusetts: Duxbury Press, 1973.
26. Meyer, Paul L. Introductory Probability and Statistical Applications. (second edition). Reading, Massachusetts: Addison-Wesley Publishing Co., 1970.
27. Mueller, Donald D., Colonel, USAF, TAC/DRFA. F-15 Peacetime LCOM Coordination Letter. Langley Air Force Base, Virginia: 23 March 1976.
28. Siegel, Sidney. Nonparametric Statistics. New York: McGraw-Hill Book Co., 1956.

29. Smith, Donald C., Colonel, USAF, TAC/XPMM. Letter to HQ AFIT/ENS Concerning a proposed thesis topic. Langley Air Force Base, Ohio: 18 December 1975.
30. TAC Syllabi F1500 B,I, and TX. F-15 Combat Crew Training. Langley Air Force Base, Ohio: April 1976.
31. Tetmeyer, Donald C., Lt. Colonel, USAF, ASD/ENECC. Personal Interview. Wright-Patterson Air Force Base, Ohio: October 1976.
32. Tetmeyer, Donald C. and Sharon R. Nichos. Simulation Maintenance Manning for New Weapon Systems: Maintenance Data Analysis Programs. AFHRL-TR-74-97 (III). Brooks Air Force Base, Texas: Headquarters, Air Force Human Resources Laboratory, May 1974.
33. Tetmeyer, Donald C. and William D. Moody. Simulating Maintenance Manning for New Weapon Systems: Building and Operating a Simulation Model. AFHRL-TR-74-97 (II). Brooks Air Force Base, Texas: Headquarters, Air Force Human Resources Laboratory, December 1974.
34. Tetmeyer, Donald C. and William D. Moody. Simulation Maintenance Manning for New Weapon Systems: Manpower Programs. AFHRL-TR-74-97 (V). Brooks Air Force Base, Texas: Headquarters, Air Force Human Resources Laboratory, April 1974.
35. Theil, Henri. Principles of Econometrics. New York: John Wiley & Sons, Inc., 1971
36. T. O. 1-F-15A-06. Work Unit Code Manual: USAF Series F-15A 72-113 and TF-15A 73-108 Aircraft. 1 December 1973.

APPENDIX A

F-15 TFTW DATA BASE

2. The networks include the following coding conventions.

a. ACTION CODES:

F - Failure Clock.

T - Troubleshoot On-aircraft.

X - Work to Facilitate Maintenance (Access, Preparation, etc.).

A - Get and Hookup Aircraft Ground Equipment.

R - Remove/Replace LRU.

M - Repair On-aircraft.

V - Verify Systems Works.

W - Check and Repair In-shop.

K - Check OK in Shop.

N - Check and NRTS/Condemn.

J - Aircraft Handling.

B - Loading/Downloading.

G - Fueling.

H - Flightline Inspection/Service.

P - Phase Inspection.

C - Call Section (Reference to a sub-network).

Q - Draw LRU from Supply.

D - Decrement a Failure Clock.

Z - Fly Sortie.

b. Task names for unscheduled maintenance use the appropriate LRU work unit code (WUC). Task names for general aircraft servicing and support tasks are mnemonic abbreviations of the task description. For example, the task of reloading the gun is named LDGUN, and the task of servicing liquid oxygen is named LOXSV.

c. LCOM SELECTION MODES:

- A - Non-Mutually Exclusive Probability.
- C - Call Section Reference to a Sub-network.
- D - Task Done in Indicated Order.
- E - Mutually Exclusive Probability.
- F - Task Done if Indicated Failure Occurs.
- G - Relative Probability, Non-Mutually Exclusive.
- H - Task Not Done if Indicated Failure Occurs.
- I - Task Done Only if Cannibalization is Required.
- X - Task Done When Failure Occurs in Subsequent Tasks.
- S - Sortie (Externally Scheduled).

3. Task resource names are the Air Force Specialty Codes (AFSC's) of maintenance personnel, or mnemonic abbreviations for ground support equipment. For example, LXCART indicates a liquid oxygen service cart, and D60 indicates an M32-60 Ground Power Unit. The fourth (skill level) digit of the Air Force Specialty Code is replaced by an X, except in cases where a further shredout is necessary. Table A-I depicts each AFSC whose manpower was estimated by LCOM, the corresponding LCOM shredout, and the work description.

Table A-I

LCOM AFSC Shredout Used in F-15 Data Base

AFSC	LCOM Shredout	Work Description
326X1D	326A1	Automatic Test Station
326X2A	326A2	Inertial Navigation System/Weapon Control
326X1C	326B1	Manual Test Station
326X2B	326B2	Automatic Flight Control/Instruments
326X2C	326C2	Communications/Navigation/Electronic Counter Measures
423X0	423X0	Electrical Systems
423X1	422X1	Environmental Systems
423X2	422X2	Egress Systems
423X3	424X0	Fuel Systems
423X4	421X2	Pneudraulics
426X2	432X0	Jet Engine Shop/Jet Engine Flight Line
	432T0	Jet Engine Test Cell
431X1	431X1	Flight Line Crew Chief
	431P1	Phase Inspection
531X3	531X3	Structural Repair

JOB	DATE	TIME	DESCRIPTION	STATUS	REMARKS	PAGE
JGP151.1275	95	09 1 001 17	MAIN NETWORKS FOR F15 TEST OPERATION	M	THIS IS AN UPDATE OF THE F15 STUDY ACCOMPLISHED OCTOBER 1975. THE CLOCKS HAVE BEEN CHANGED TO REFLECT ACTUAL LOGS FROM THE PERIOD SEP 75 THRU FEB 76. THE PHASE AND ENGINE NETWORKS HAVE BEEN CHANGED DRASTICALLY. THE PHASE AND LOGS AND AFLO NO LONGER SUPPORT THE NEW 24 HOUR PHASE INSPECTION THAT WAS SUGGESTED BY SGT JONES. THE ENGINE HAD TO BE REWORKED TO REFLECT THIS. ALL NETWORK CHANGES WERE ACCOMPLISHED UNDER THE DIRECTION OF A TAG REPRESENTATIVE AND ALL SIMULATION WILL BE DONE AS AN AFIT THIS TOPIC.	09ASE
			PREFLIGHT DUMMY SORTIE	M		09ASE
J001	Z0000			S	00010 11	09ASE 2
J001	00000			S	00010 31	09ASE 2
CALLRE	00000	J0004		D	00010 31	09ASE 2
J001	00000	J0006		D	00010 31	09ASE 2
J001	00000			D	00010 31	09ASE 2
J001	00000			D	00010 31	09ASE 2
J001	00000			D	00010 31	09ASE 2
CALLRE	00000			A	30 00010 31	09ASE 2
CALLRE	00000			D	00010 31	09ASE 27
CALLRE	00000			M	00010	09ASE 28
CALLRE	00000	J0013		S	1 00010 31	09ASE 29
J001	00000	J0012		D	00010 31	09ASE 30
J001	00000			D	00010 31	09ASE 31
J001	00000			D	00010 31	09ASE 32
J001	00000			D	00010 31	09ASE 33
CLG001	00000			A	17 00010 31	09ASE 34
CLG001	00000			A	34 00010 31	09ASE 35
CALLRE	00000			M	00010	09ASE 36
CALLRE	00000			M	00010	09ASE 37
CALLRE	00000			M	00010	09ASE 38
CALLRE	00000			M	00010	09ASE 39
CALLRE	00000			M	00010	09ASE 40
CALLRE	00000			M	00010	09ASE 41
CALLRE	00000			M	00010	09ASE 42
CALLRE	00000			M	00010	09ASE 43
AA1	CALLRE	AA2		D	00010 31	09ASE 44
AA2	CALLRE	AA3		D	00010 31	09ASE 45
AA3	CALLRE	AA4		D	00010 31	09ASE 46
AA4	CALLRE	AA5		D	00010 31	09ASE 47
AA5	CALLRE	AA6		D	00010 31	09ASE 48
AA6	CALLRE	AA7		D	00010 31	09ASE 49
AA7	CALLRE	AA8		D	00010 31	09ASE 50
AA8	CALLRE	AA9		D	00010 31	09ASE 51
AA9	CALLRE	AA10		D	00010 31	09ASE 52
AA10	CALLRE	AA11		D	00010 31	09ASE 53
AA11	CALLRE	AA12		D	00010 31	09ASE 54
AA12	CALLRE	AA13		D	00010 31	09ASE 55
AA13	CALLRE	AA14		D	00010 31	09ASE 56
AA14	CALLRE	AA15		D	00010 31	09ASE 57
AA15	CALLRE	AA16		D	00010 31	09ASE 58
AA16	CALLRE	AA17		D	00010 31	09ASE 59
AA17	CALLRE	AA18		D	00010 31	09ASE 60
AA18	CALLRE	AA19		D	00010 31	09ASE 61
AA19	CALLRE	AA20		D	00010 31	09ASE 62
AA20	CALLRE	AA21		D	00010 31	09ASE 63

AA18	CALLS: AA19	C	00010 31		DBASE	64
AA19	CLGUMI	C	00010 31		DBASE	65
		4	00010 31	AIR-AIR TNG TURN TO TURN	DBASE	66
AA21	HLUNCH AA22	C	00010 31		DBASE	67
AA22	HSTRTE AA23	C	00010 31		DBASE	68
AA23	JCRMT1 AA24	C	00010 31		DBASE	69
AA24	DALEGE2 AA25	C	00010 31		DBASE	70
AA25	CALJK2 AA26	C	00010 31		DBASE	71
AA26	CALLOM AA27	C	00010 31		DBASE	72
AA27	CALLS: AA28	C	00010 31		DBASE	73
AA28	JTAXI AA29	C	00010 31		DBASE	74
AA29	HECPG2 AA30	C	00010 31		DBASE	75
AA30	ZCRMT2 AA31	S	00010 31		DBASE	76
AA31	HEORTM AA32	C	00010 31		DBASE	77
AA32	HTAXI AA33	C	00010 31		DBASE	78
AA33	JCRMT2	C	00010 31		DBASE	79
AA34	JCRMT5	C	00010 31		DBASE	80
AA35	GASFL5 AA36	C	00010 31		DBASE	81
AA36	HTHPU2	C	00010 31		DBASE	82
AA37	HTHPU2	C	00010 31		DBASE	83
AA38	DALEGE1 AA39	C	00010 31		DBASE	84
AA39	CALJK1 AA40	C	00010 31		DBASE	85
AA40	CALLES1 AA41	C	00010 31		DBASE	86
AA41	CLGUMI	C	00010 31		DBASE	87
		4	00010 31	AIR-AIR TNG TURN TO LAST	DBASE	88
AA42	HLUNCH AA43	C	00010 31		DBASE	89
AA43	HSTRTE AA44	C	00010 31		DBASE	90
AA44	JCRMT1 AA45	C	00010 31		DBASE	91
AA45	DALEGE2 AA46	C	00010 31		DBASE	92
AA46	CALJK2 AA47	C	00010 31		DBASE	93
AA47	CALLOM AA48	C	00010 31		DBASE	94
AA48	CALLS: AA49	C	00010 31		DBASE	95
AA49	JTAXI AA50	C	00010 31		DBASE	96
AA50	HECPG2 AA51	C	00010 31		DBASE	97
AA51	ZCRMT2 AA52	S	00010 31		DBASE	98
AA52	HEORTM AA53	C	00010 31		DBASE	99
AA53	JCRMT2	C	00010 31		DBASE	100
AA54	JCRMT5	C	00010 31		DBASE	101
AA55	HTAXI AA56	C	00010 31		DBASE	102
AA56	GASFL5 AA57	C	00010 31		DBASE	103
AA57	DARM AA58	C	00010 31	3 29L 1-6220	DBASE	104
AA58	MPJST2	C	00010 31	10 29L 1-31X1	DBASE	105
AA59	MPJST2	C	00010 31	10 29L 1-31X2	DBASE	106
AA60	DALEGE1 AA59	C	00010 31		DBASE	107
AA61	CALJK1 AA60	C	00010 31		DBASE	108
AA62	CALLS: AA61	C	00010 31		DBASE	109
AA63	CLGUMI	C	00010 31		DBASE	110
		4	00010 31	AIR-AIR FIRST TO LAST	DBASE	111
AA101	CALPRE AA102	C	00010 31		DBASE	112
AA102	DARM AA103	C	00010 31		DBASE	113
AA103	HLUNCH AA104	C	00010 31		DBASE	114
AA104	HSTRTE AA105	C	00010 31		DBASE	115
AA105	JCRMT1 AA106	C	00010 31		DBASE	116
AA106	DALEGE2 AA107	C	00010 31		DBASE	117
AA107	CALJK2 AA108	C	00010 31		DBASE	118
AA108	CALLOM AA109	C	00010 31		DBASE	119
AA109	CALLS: AA110	C	00010 31		DBASE	120
AA110	JTAXI AA111	C	00010 31		DBASE	121
AA111	HECPG2 AA112	C	00010 31		DBASE	122
AA112	ZCRMT2 AA113	S	00010 31		DBASE	123
AA113	HEORTM AA114	C	00010 31		DBASE	124
AA114	JCRMT2	C	00010 31		DBASE	125
AA115	JCRMT5	C	00010 31		DBASE	126
AA116	HTAXI AA115	C	00010 31		DBASE	127
AA117	GASFL5 AA116	C	00010 31		DBASE	128
AA118	DARM AA117	C	00010 31		DBASE	129

AA117	WPOST:	0	00010	31		0BASE	130	
AA117	WPOST2	0	00010	31		0BASE	131	
AA117	CALEG1	AA118	0	00010	31	0BASE	132	
AA118	CALLS1	AA119	0	00010	31	0BASE	133	
AA119	CALLS:	AA120	0	00010	31	0BASE	134	
AA120	CALLS2		0	00010	31	0BASE	135	
		4	00010		CONVERSION TNG MISSION FIRST TO TURN	0BASE	136	
CON1	CALEG2	CON2	0	00010	31	0BASE	137	
CON2	HLUNCH	CON3	0	00010	31	0BASE	138	
CON3	HSTRT1	CON4	E	25	00010	31	10 29L 243141	
CON3	HSTRT2	CON4	E	75	00010	31		
CON4	DEMT1		0	00010	31	0BASE	141	
CON4	CALEG2	CON5	0	00010	31	0BASE	142	
CON5	CALLK2	CON6	0	00010	31	0BASE	143	
CON6	CALLOM	CON7	0	00010	31	0BASE	144	
CON7	CALLS1	CON8	0	00010	31	0BASE	145	
CON8	JTAX1	CON9	0	00010	31	2 28L	0BASE	146
CON9	HEPFGD	CON10	0	00010	31	0BASE	147	
CON10	ZALLES1	CON11	E	00010	11	0BASE	148	
CON11	HECPIN	CON12	0	00010	31	0BASE	149	
CON12	DEMT2		0	00010	31	0BASE	151	
CON12	HTAX1	CON13	0	00010	31	0BASE	151	
CON13	GASFL5	CON14	0	00010	31	0BASE	152	
CON14	HTHRU1		0	00010	31	0BASE	153	
CON14	HTHRU2		0	00010	31	0BASE	154	
CON15	CALEG1	CON15	0	00010	31	0BASE	155	
CON15	CALLK1	CON16	0	00010	31	0BASE	156	
CON15	CALLS1		0	00010	31	0BASE	157	
		4	00010		CONVERSION TNG MISSION TURN TO LAST	0BASE	158	
CON31	HLUNCH	CON31	0	00010	31	0BASE	159	
CON31	HSTRT1	CON32	E	25	00010	31	0BASE	160
CON31	HSTRT2	CON32	E	75	00010	31	0BASE	161
CON32	DEMT1		0	00010	31	0BASE	162	
CON32	CALEG2	CON33	0	00010	31	0BASE	163	
CON33	CALLK2	CON34	0	00010	31	0BASE	164	
CON34	CALLOM	CON35	0	00010	31	0BASE	165	
CON35	CALLS1	CON36	0	00010	31	0BASE	166	
CON35	JTAX1	CON37	0	00010	31	0BASE	167	
CON37	HEPFGD	CON38	0	00010	31	0BASE	168	
CON38	ZALLES1	CON39	E	00010	11	0BASE	169	
CON39	HECPIN	CON40	0	00010	31	0BASE	170	
CON39	DEMT2		0	00010	31	0BASE	171	
CON40	HTAX1	CON41	0	00010	31	0BASE	172	
CON41	GASFL5	CON42	0	00010	31	0BASE	173	
CON42	WPOST1		0	00010	31	0BASE	174	
CON42	WPOST2		0	00010	31	0BASE	175	
CON43	CALEG1	CON43	0	00010	31	0BASE	176	
CON43	CALLK1	CON44	0	00010	31	0BASE	177	
CON44	CALLS1		0	00010	31	0BASE	178	
		4	00010		CONVERSION TNG MISSION TURN TO TURN	0BASE	179	
CON51	HLUNCH	CON51	0	00010	31	0BASE	180	
CON51	HSTRT1	CON52	E	25	00010	31	0BASE	181
CON51	HSTRT2	CON52	E	75	00010	31	0BASE	182
CON52	DEMT1		0	00010	31	0BASE	183	
CON52	CALEG2	CON53	0	00010	31	0BASE	184	
CON53	CALLK2	CON54	0	00010	31	0BASE	185	
CON54	CALLOM	CON55	0	00010	31	0BASE	186	
CON55	CALLS1	CON56	0	00010	31	0BASE	187	
CON56	JTAX1	CON57	0	00010	31	0BASE	188	
CON57	HEPFGD	CON58	0	00010	31	0BASE	189	
CON58	ZALLES1	CON59	E	00010	11	0BASE	190	
CON59	HECPIN	CON60	0	00010	31	0BASE	191	
CON59	DEMT2		0	00010	31	0BASE	192	
CON60	HTAX1	CON61	0	00010	31	0BASE	193	
CON61	GASFL5	CON62	0	00010	31	0BASE	194	
CON62	HTHRU1		0	00010	31	0BASE	195	

CON62	MTMPU2	0	00010	31			DBASE	196
CON62	CALLEG	CON62	0	00010	31		DBASE	197
CON63	CALJJK	CON66	0	00010	31		DBASE	198
CON64	CALLES	0	00010	31			DBASE	199
CON101	CALPPE	CON101	0	00010	31		DBASE	200
CON102	HLUNGH	CON102	0	00010	31		DBASE	201
CON102	HSTAT2	CON103	E	75	00010	31	DBASE	202
CON102	HSTAT1	CON103	E	25	00010	31	DBASE	203
CON103	DCRMT1	0	00010	31			DBASE	204
CON103	CALLEG	CON104	0	00010	31		DBASE	205
CON104	CALJJK2	CON105	0	00010	31		DBASE	206
CON105	CALJJK	CON106	0	00010	31		DBASE	207
CON106	CALLES	CON107	0	00010	31		DBASE	208
CON107	HSTAT	CON108	0	00010	31		DBASE	209
CON108	MEORGD	CON109	0	00010	31		DBASE	210
CON109	Z...	CON110	0	00010	31		DBASE	211
CON110	MEORIN	CON111	0	00010	31		DBASE	212
CON111	DCRMT2	0	00010	31			DBASE	213
CON111	HSTAT	CON112	0	00010	31		DBASE	214
CON112	GASFI5	CON113	0	00010	31		DBASE	215
CON113	HPOST1	0	00010	31			DBASE	216
CON113	HPOST2	0	00010	31			DBASE	217
CON113	CALLEG	CON114	0	00010	31		DBASE	218
CON114	CALLES	CON115	0	00010	31		DBASE	219
CON115	CALLES	0	00010	31			DBASE	220
						SCHEDULED WASH 900AY	DBASE	221
JL103	Z...	JL104	S	00010	11		DBASE	222
JL103	DCRMT2	0	00010	31			DBASE	223
JL104	JTOWAD	JL105	0	00010	32	4 29L 5-31X1	DBASE	224
JL105	JTOWAD	JL106	0	00010	32	10 29L 1431X1 1TJG	DBASE	225
JL106	JTOWAD	JL107	0	00010	32	30 29L 3431X1 1WRACK	DBASE	226
JL107	JLUB	0	00010	32	25 29L 1531X1		DBASE	227
JL107	JLUB	0	00010	32	10 29L 1-31X1		DBASE	228
JL108	JTOWAD	0	00010	32			DBASE	229
TANK1	DOTANK	TANK1	F	1337	OTANK	HANG CENTERLINE EXTERNAL TANK	DBASE	230
TANK1	DMANGT	TANK2	0		OTANK	31	DBASE	231
TANK1	DMANGT	TANK2	0		OTANK	31	DBASE	232
TANK2	DMANGT	TANK2	0		OTANK	31	DBASE	233
TANK3	CALGFI	TANK4	0		OTANK	31	DBASE	234
TANK4	JTOWAD	TANK5	0		OTANK	31	DBASE	235
TANK3	GASFI5		0		OTANK	31	DBASE	236
						F-15 PHASE INFO FROM TSST ATKINS AND	DBASE	237
						SSGT SOUND AT LUKE AFB ARIZ APRIL 75	DBASE	238
						ALL PHASES INCLUDE A FIX PHASE WHERE	DBASE	239
						MINOR MAINTENANCE IS PERFORMED, AND	DBASE	240
						PHASE 1 & 4 INCLUDE A NOI OF THE RAMP	DBASE	241
						I/A/W MCI66-28.	DBASE	242
PHASE	DCRMT	PHASE	0		PHASE	32	DBASE	243
PHASE	DCRMT	0			PHASE	32	DBASE	244
PHASE	DCRMT	PHASE	0		PHASE	32	DBASE	245
PHASE	DCRMT	0			PHASE	32	DBASE	246
PHASE	OTANK	PHASE	0		PHASE	32	DBASE	247
PHASE	HANGAR	PHASE	0		PHASE	32	DBASE	248
PHASE	Z...	PHASE	0		PHASE	11	DBASE	249
PHASE	FPHASE	PHASE	E	17	PHASE	32	DBASE	250
PHASE	FPHASE	PHASE	E	17	PHASE	32	DBASE	251
PHASE	FPHASE	PHASE	E	17	PHASE	32	DBASE	252
PHASE	FPHASE	PHASE	E	16	PHASE	32	DBASE	253
PHASE	FPHASE	PHASE	E	17	PHASE	32	DBASE	254
PHASE	FPHASE	PHASE	E	16	PHASE	32	DBASE	255
PHASE	PHASE	PHASE	0		PHASE	32	DBASE	256
PHASE	PHASE	PHASE	0		PHASE	32	DBASE	257
PHASE	PHASE	PHASE	0		PHASE	32	DBASE	258
PHASE	PHASE	PHASE	0		PHASE	32	DBASE	259
PHASE	PHASE	PHASE	0		PHASE	32	DBASE	260
PHASE	PHASE	PHASE	0		PHASE	32	DBASE	261

PH4.16	SPFC1	PH4.17	C	PHASE 32		OBASE 262
PH4.17	PJUM1	PH4.18	E	9. PHASE 32		OBASE 263
PH4.17	PQC	PH4.19	E	1. PHASE 32	10 29L 4431P1	OBASE 264
PH4.18	CFIX	PH4.20	C	PHASE 32		OBASE 265
PH4.19	CPAN	PH4.21	C	PHASE 32		OBASE 266
PH4.20	PTDM	PH4.22	C	PHASE 32	6 29L 5431P1	OBASE 267
PH4.21	PYDI	PH4.23	C	PHASE 32	45 29L 2531X5	OBASE 268
PH4.22	PEJ	PH4.24	E	3. PHASE 32	21 29L 2431P1	OBASE 269
PH4.22	PJUM1	PH4.25	E	55. PHASE 32		OBASE 270
PH4.23	PTCWO		C	PHASE 32	5 29L 5431P1	OBASE 271
CPAN	PPAN1		C	PHASE 32	10 29L 1431P1	OBASE 272
CPAN	PPAN2		C	PHASE 32	10 29L 1431P1	OBASE 273
CPAN	PPAN3		C	PHASE 32	10 29L 1431P1	OBASE 274
CPAN	PPAN4		C	PHASE 32	10 29L 1431P1	OBASE 275
CSPEC1	LOOK1		C	PHASE 32		OBASE 276
CSPEC1	LOOK1		C	PHASE 32		OBASE 277
CSPEC1	PINST1		C	PHASE 32	20 29L 232632	OBASE 278
CSPEC1	PE/??		C	PHASE 32	7 29L 232612	OBASE 279
CSPEC1	PSTOUT		E	5. PHASE 32	20 29L 2431P1	OBASE 280
CSPEC1	PE/??		E	5. PHASE 32	10 29L 1431P1	OBASE 281
CLU1	LJ11		C	PHASE 32	1 29L 2431P1	OBASE 282
CLU1	LJ12		C	PHASE 32	3 29L 2431P1	OBASE 283
CLU1	LJ11		C	PHASE 32		OBASE 284
CLU1	LJ12		C	PHASE 32		OBASE 285
CLU1	LJ11		C	PHASE 32		OBASE 286
CLU1	LJ12		C	PHASE 32		OBASE 287
CLU1	LJ11		C	PHASE 32		OBASE 288
CLU1	LJ12		C	PHASE 32		OBASE 289
CLU1	LJ11		C	PHASE 32		OBASE 290
CLU1	LJ12		C	PHASE 32		OBASE 291
CLU1	LJ11		C	PHASE 32		OBASE 292
CLU1	LJ12		C	PHASE 32		OBASE 293
CLU1	LJ11		C	PHASE 32		OBASE 294
CLU1	LJ12		C	PHASE 32		OBASE 295
CLU1	LJ11		C	PHASE 32		OBASE 296
CLU1	LJ12		C	PHASE 32		OBASE 297
CLU1	LJ11		C	PHASE 32		OBASE 298
CLU1	LJ12		C	PHASE 32		OBASE 299
CLU1	LJ11		C	PHASE 32		OBASE 300
CLU1	LJ12		C	PHASE 32		OBASE 301
CLU1	LJ11		C	PHASE 32		OBASE 302
CLU1	LJ12		C	PHASE 32		OBASE 303
CLU1	LJ11		C	PHASE 32		OBASE 304
CLU1	LJ12		C	PHASE 32		OBASE 305
CFIX	PFIX1		C	PHASE 32	20 29L 4431P1	OBASE 306
CFIX	PFIX1		C	PHASE 32	10 29L 2421X2	OBASE 307
CFIX	PFIX5		C	PHASE 32	30 29L 1531X3	OBASE 308
LOOK1	LOOK1		C	PHASE 32	30 29L 1431P1	OBASE 309
LOOK1	LOOK2		C	PHASE 32	33 29L 1431P1	OBASE 310
LOOK1	LOOK2		C	PHASE 32		OBASE 311
LOOK1	LOOK2		C	PHASE 32	2 29L 1431P1	OBASE 312
LOOK1	LOOK1		C	PHASE 32		OBASE 313
LOOK1	LOOK1		C	PHASE 32		OBASE 314
LOOK1	LOOK2		C	PHASE 32		OBASE 315
LOOK1	LOOK-		C	PHASE 32	5 29L 1431P1	OBASE 316
LOOK1	LOOK-		C	PHASE 32		OBASE 317
LOOK1	LOOK5		C	PHASE 32	1 29L 1431P1	OBASE 318
LOOK1	LOOK5		C	PHASE 32		OBASE 319
PH4.25	PS4FE	PH4.26	C	PHASE 32		OBASE 320
PH4.26	CPAN	PH4.27	C	PHASE 32		OBASE 321
PH4.26	PH4.27	PH4.28	C	PHASE 32	40 29L 1431P1 2421X2	OBASE 322
PH4.27	CSPEC2	PH4.28	C	PHASE 32		OBASE 323
PH4.28	PJUM1	PH4.29	E	9. PHASE 32		OBASE 324
PH4.28	PJC	PH4.30	E	10. PHASE 32		OBASE 325
PH4.29	CFIX	PH4.31	C	PHASE 32		OBASE 326
PH4.30	CPAN	PH4.32	C	PHASE 32		OBASE 327

PH0016 T3H01T	0	PHASE 32	5 29L 3431X1	DBASE	328
CSPE02 BT1ST2	0	PHASE 32	17 29L 312532	DBASE	329
CSPE02 PF1C1	0	PHASE 32		DBASE	330
CSPE02 PHEAD2 PH1123	0	PHASE 32	11 29L 446273	DBASE	331
PH0019 SNEAP1 PH0013	0	PHASE 32	107 29L 2-02X1	DBASE	332
PH0017 PHEAD2	0	PHASE 32	20 29L 440273	DBASE	333
CSPE02 CL00K2	0	PHASE 32		DBASE	334
CSPE02 PHEAD2	0	PHASE 32		DBASE	335
CSPE02 PHV022	0	PHASE 32	15 29L 1421X2	DBASE	336
CSPE02 P1FS	0	PHASE 32	11 29L 1432Y1	DBASE	337
CSPE02 CL032	0	PHASE 32		DBASE	339
CLU02 LU31	0	PHASE 32		DBASE	339
CLU02 LU31	0	PHASE 32		DBASE	340
CLU02 LU31	0	PHASE 32		DBASE	341
CLU02 LU31	0	PHASE 32		DBASE	342
CLU02 LU31	0	PHASE 32		DBASE	343
CLU02 LU31	0	PHASE 32		DBASE	344
CLU02 LU31	0	PHASE 32		DBASE	345
CLU02 LU31	0	PHASE 32		DBASE	346
CLU02 LU31	0	PHASE 32		DBASE	347
CLU02 LU31	0	PHASE 32		DBASE	348
CLU02 LU31	0	PHASE 32		DBASE	349
CLU02 LU31	0	PHASE 32		DBASE	350
CLU02 LU31	0	PHASE 32		DBASE	351
CLU02 LU31	0	PHASE 32		DBASE	352
CLU02 LU31	0	PHASE 32		DBASE	353
CLU02 LU31	0	PHASE 32		DBASE	354
CLU02 LU31	0	PHASE 32		DBASE	355
CLU02 LU31	0	PHASE 32		DBASE	356
CLU02 LU31	0	PHASE 32		DBASE	357
CLU02 LU31	0	PHASE 32		DBASE	358
CLU02 LU31	0	PHASE 32		DBASE	359
CLU02 LU31	0	PHASE 32		DBASE	360
CLU02 LU31	0	PHASE 32		DBASE	361
CLU02 LU31	0	PHASE 32		DBASE	362
CLU02 LU31	0	PHASE 32		DBASE	363
CSPE02 PHEAD2 PH0013	0	PHASE 32	10 29L 3-0270	DBASE	364
PH0015 SNEAP2 PH0013	0	PHASE 32	40 29L 2-02X1	DBASE	365
PH0011 PHEAD2	0	PHASE 32	10 29L 3-0270	DBASE	366
CSPE02 CL00K3	0	PHASE 32		DBASE	367
CSPE02 CL00K3	0	PHASE 32		DBASE	368
CSPE02 PHEAD2	0	PHASE 32	30 29L 1432X0	DBASE	369
CSPE02 P1UM1 PH0011	0	PHASE 32		DBASE	370
PH0011 PHEAD2	0	PHASE 32	5 29L 1-21X0	DBASE	371
PH0011 PHEAD2	0	PHASE 32	20 29L 1-21X0	DBASE	372
CSPE02 PHEAD2	0	PHASE 32	40 29L 2-02X1	DBASE	373
CLU02 LU31	0	PHASE 32		DBASE	374
CLU02 LU31	0	PHASE 32		DBASE	375
CLU02 LU31	0	PHASE 32		DBASE	376
CLU02 LU31	0	PHASE 32		DBASE	377
CLU02 LU31	0	PHASE 32		DBASE	378
CLU02 LU31	0	PHASE 32		DBASE	379
CLU02 LU31	0	PHASE 32		DBASE	380
CLU02 LU31	0	PHASE 32		DBASE	381
CLU02 LU31	0	PHASE 32		DBASE	382
CLU02 LU31	0	PHASE 32		DBASE	383
CLU02 LU31	0	PHASE 32		DBASE	384
CLU02 LU31	0	PHASE 32		DBASE	385
CLU02 LU31	0	PHASE 32		DBASE	386
CLU02 LU31	0	PHASE 32		DBASE	387
CLU02 LU31	0	PHASE 32		DBASE	388
CLU02 LU31	0	PHASE 32		DBASE	389
CLU02 LU31	0	PHASE 32		DBASE	390
CLU02 LU31	0	PHASE 32		DBASE	391
CLU02 LU31	0	PHASE 32		DBASE	392
CLU02 LU31	0	PHASE 32		DBASE	393
PH0012 CPAN PH0013	0	PHASE 32		DBASE	394
PH0013 PHEAD2 PH0014	0	PHASE 32		DBASE	395

PH0101	PH0101	PH0101	0	PHASE 32	10 29L 343101	ORASE 394
PH0102	PH0102	PH0102	0	PHASE 32		ORASE 395
PH0103	PH0103	PH0103	0	PHASE 32		ORASE 396
PH0104	PH0104	PH0104	0	PHASE 32		ORASE 397
PH0105	PH0105	PH0105	0	PHASE 32		ORASE 398
PH0106	PH0106	PH0106	0	PHASE 32		ORASE 399
PH0107	PH0107	PH0107	0	PHASE 32		ORASE 400
PH0108	PH0108	PH0108	0	PHASE 32		ORASE 401
PH0109	PH0109	PH0109	0	PHASE 32		ORASE 402
PH0110	PH0110	PH0110	0	PHASE 32		ORASE 403
PH0111	PH0111	PH0111	0	PHASE 32		ORASE 404
PH0112	PH0112	PH0112	0	PHASE 32	10 29L 1422X1	ORASE 405
PH0113	PH0113	PH0113	0	PHASE 32	10 29L 1421X2	ORASE 406
PH0114	PH0114	PH0114	0	PHASE 32	10 29L 1423X0	ORASE 407
PH0115	PH0115	PH0115	0	PHASE 32	10 29L 1432X0	ORASE 408
PH0116	PH0116	PH0116	0	PHASE 32		ORASE 409
PH0117	PH0117	PH0117	0	PHASE 32		ORASE 410
PH0118	PH0118	PH0118	0	PHASE 32		ORASE 411
PH0119	PH0119	PH0119	0	PHASE 32		ORASE 412
PH0120	PH0120	PH0120	0	PHASE 32		ORASE 413
PH0121	PH0121	PH0121	0	PHASE 32		ORASE 414
PH0122	PH0122	PH0122	0	PHASE 32		ORASE 415
PH0123	PH0123	PH0123	0	PHASE 32		ORASE 416
PH0124	PH0124	PH0124	0	PHASE 32		ORASE 417
PH0125	PH0125	PH0125	0	PHASE 32		ORASE 418
PH0126	PH0126	PH0126	0	PHASE 32		ORASE 419
PH0127	PH0127	PH0127	0	PHASE 32		ORASE 420
PH0128	PH0128	PH0128	0	PHASE 32		ORASE 421
PH0129	PH0129	PH0129	0	PHASE 32		ORASE 422
PH0130	PH0130	PH0130	0	PHASE 32		ORASE 423
PH0131	PH0131	PH0131	0	PHASE 32		ORASE 424
PH0132	PH0132	PH0132	0	PHASE 32		ORASE 425
PH0133	PH0133	PH0133	0	PHASE 32		ORASE 426
PH0134	PH0134	PH0134	0	PHASE 32		ORASE 427
PH0135	PH0135	PH0135	0	PHASE 32		ORASE 428
PH0136	PH0136	PH0136	0	PHASE 32		ORASE 429
PH0137	PH0137	PH0137	0	PHASE 32		ORASE 430
PH0138	PH0138	PH0138	0	PHASE 32		ORASE 431
PH0139	PH0139	PH0139	0	PHASE 32		ORASE 432
PH0140	PH0140	PH0140	0	PHASE 32	35 29L 2422X2	ORASE 433
PH0141	PH0141	PH0141	0	PHASE 32	30 29L 2422X2	ORASE 434
PH0142	PH0142	PH0142	0	PHASE 32	5 29L 1422X2	ORASE 435
PH0143	PH0143	PH0143	0	PHASE 32	35 29L 2422X2	ORASE 436
PH0144	PH0144	PH0144	0	PHASE 32	35 29L 1431P1	ORASE 437
PH0145	PH0145	PH0145	0	PHASE 32	15 29L 2422X2	ORASE 438
PH0146	PH0146	PH0146	0	PHASE 32	15 29L 2422X2	ORASE 439
PH0147	PH0147	PH0147	0	PHASE 32	2 29L 1422X2	ORASE 440
PH0148	PH0148	PH0148	0	PHASE 32	15 29L 2422X2	ORASE 441
PH0149	PH0149	PH0149	0	PHASE 32	10 29L 1431P1	ORASE 442
PH0150	PH0150	PH0150	0	PHASE 32		ORASE 443
PH0151	PH0151	PH0151	0	PHASE 32		ORASE 444
PH0152	PH0152	PH0152	0	PHASE 32		ORASE 445
PH0153	PH0153	PH0153	0	PHASE 32	10 29L 2421X2	ORASE 446
PH0154	PH0154	PH0154	0	PHASE 32		ORASE 447
PH0155	PH0155	PH0155	0	PHASE 32		ORASE 448
PH0156	PH0156	PH0156	0	PHASE 32		ORASE 449
PH0157	PH0157	PH0157	0	PHASE 32		ORASE 450
PH0158	PH0158	PH0158	0	PHASE 32		ORASE 451
PH0159	PH0159	PH0159	0	PHASE 32		ORASE 452
PH0160	PH0160	PH0160	0	PHASE 32		ORASE 453
PH0161	PH0161	PH0161	0	PHASE 32		ORASE 454
PH0162	PH0162	PH0162	0	PHASE 32		ORASE 455
PH0163	PH0163	PH0163	0	PHASE 32		ORASE 456
PH0164	PH0164	PH0164	0	PHASE 32		ORASE 457
PH0165	PH0165	PH0165	0	PHASE 32		ORASE 458
PH0166	PH0166	PH0166	0	PHASE 32		ORASE 459

S1460	CALYU	S1465	3	S1460	31		OBASE	461
S1465	VS1462	VS1462	3	S1465	31	25 29L 2-62X1 1067 1TU22A	OBASE	462
S231	VS231	VS231	4	S231	31	54R 40RSCOPE 3TWN PHASES 15 77	OBASE	463
S231A	VS231A	VS231A	0	S231A	31	40 29L 2432X1	OBASE	464
S234	VS234	VS234	4	S234	31	MAIN FUEL PUMP TTY 15 77	OBASE	465
S234A	VS234A	VS234A	49	S234A	31		OBASE	466
S271	VS271	VS271	0	S271	31	75 29L 2432X1	OBASE	467
S274	VS274	VS274	0	S274	31	5 29L 3-32X1	OBASE	468
S471	VS471	VS471	6	S471	31	10X 6L 2AY 2AK 2KX	OBASE	469
S471A	VS471A	VS471A	0	S471A	31	5 29L 3-32X1	OBASE	470
S751	VS751	VS751	3	S751	31	PACKS/LAUNCHERS TOI WHEN FIRED	OBASE	471
S751A	VS751A	VS751A	0	S751A	31		OBASE	472
S752	VS752	VS752	0	S752	31		OBASE	473
S752A	VS752A	VS752A	0	S752A	31		OBASE	474
S753	VS753	VS753	0	S753	31		OBASE	475
S753A	VS753A	VS753A	0	S753A	31		OBASE	476
S754	VS754	VS754	16	S754	31	ARM SYS OK WITH GP TEST 50	OBASE	477
S754A	VS754A	VS754A	0	S754A	31	25 29L 3-62X1 1067 1TU22A	OBASE	478
JGUN1	VSJGUN1	VSJGUN1	32	JGUN1	32	F15 GUN SCHEDULED INSPECTIONS	OBASE	479
JGUN2	VSJGUN2	VSJGUN2	32	JGUN2	32		OBASE	480
JGUN3	VSJGUN3	VSJGUN3	32	JGUN3	32		OBASE	481
JGUN4	VSJGUN4	VSJGUN4	32	JGUN4	32		OBASE	482
JGUN5	VSJGUN5	VSJGUN5	32	JGUN5	32		OBASE	483
JGUN6	VSJGUN6	VSJGUN6	32	JGUN6	32		OBASE	484
JGUN7	VSJGUN7	VSJGUN7	32	JGUN7	32		OBASE	485
JGUN8	VSJGUN8	VSJGUN8	32	JGUN8	32		OBASE	486
JGUN9	VSJGUN9	VSJGUN9	32	JGUN9	32		OBASE	487
JGUN10	VSJGUN10	VSJGUN10	32	JGUN10	32		OBASE	488
JGUN11	VSJGUN11	VSJGUN11	32	JGUN11	32		OBASE	489
JGUN12	VSJGUN12	VSJGUN12	32	JGUN12	32		OBASE	490
JGUN13	VSJGUN13	VSJGUN13	32	JGUN13	32		OBASE	491
JGUN14	VSJGUN14	VSJGUN14	32	JGUN14	32		OBASE	492
JGUN15	VSJGUN15	VSJGUN15	32	JGUN15	32		OBASE	493
11A00	VS11A00	VS11A00	21	11A00	21	FWD FUS	OBASE	494
11A01	VS11A01	VS11A01	21	11A01	21	F4E 11A1 3 0 0 DATA USED	OBASE	495
11A02	VS11A02	VS11A02	21	11A02	21	M50 DIESEL LUBE PM 2783 VERIFIED	OBASE	496
11A03	VS11A03	VS11A03	21	11A03	21	EXPECT OVERALL 15PM AS F4E	OBASE	497
11A04	VS11A04	VS11A04	21	11A04	21	SCREWS BIND DUE TO OVER TORQUE	OBASE	498
11A05	VS11A05	VS11A05	21	11A05	21	PANELS CRACKING	OBASE	499
11A06	VS11A06	VS11A06	21	11A06	21	BOOM CAP FARMING CRACKS ON LANDINGS	OBASE	500
11A07	VS11A07	VS11A07	21	11A07	21	CONSTANT REPAIR 8HR 200	OBASE	501
11A08	VS11A08	VS11A08	21	11A08	21	F15 MSBMA 22	OBASE	502
11A09	VS11A09	VS11A09	21	11A09	21	FWD FUSELAGE F15	OBASE	503
11A10	VS11A10	VS11A10	21	11A10	21		OBASE	504
11A11	VS11A11	VS11A11	21	11A11	21		OBASE	505
11A12	VS11A12	VS11A12	21	11A12	21		OBASE	506
11A13	VS11A13	VS11A13	21	11A13	21		OBASE	507
11A14	VS11A14	VS11A14	21	11A14	21		OBASE	508
11A15	VS11A15	VS11A15	21	11A15	21		OBASE	509
11A16	VS11A16	VS11A16	21	11A16	21		OBASE	510
11A17	VS11A17	VS11A17	21	11A17	21		OBASE	511
11A18	VS11A18	VS11A18	21	11A18	21		OBASE	512
11A19	VS11A19	VS11A19	21	11A19	21		OBASE	513
11A20	VS11A20	VS11A20	21	11A20	21		OBASE	514
11A21	VS11A21	VS11A21	21	11A21	21		OBASE	515
11A22	VS11A22	VS11A22	21	11A22	21		OBASE	516
11A23	VS11A23	VS11A23	21	11A23	21		OBASE	517
11A24	VS11A24	VS11A24	21	11A24	21		OBASE	518
11A25	VS11A25	VS11A25	21	11A25	21		OBASE	519
11A26	VS11A26	VS11A26	21	11A26	21		OBASE	520
11A27	VS11A27	VS11A27	21	11A27	21		OBASE	521
11A28	VS11A28	VS11A28	21	11A28	21		OBASE	522
11A29	VS11A29	VS11A29	21	11A29	21		OBASE	523
11A30	VS11A30	VS11A30	21	11A30	21		OBASE	524
11A31	VS11A31	VS11A31	21	11A31	21		OBASE	525

A1001	F11000	A1001	F	6	11000	21			DRASE	524
A1001	F11001	A1002	E	02	11000	21	22	29L 1431X1	DRASE	527
A1001	F11002		E	17	11000	21	12	29L 1431X1	DRASE	528
A1001	F11003		E	55	11000	21	14	29L 2531X3	DRASE	529
A1001	F11004		E	40	11000	21	10	29L 1531X0	DRASE	530
A1001	S40P	S41000	D		11000	23			DRASE	531
S41001	L11001	I11001	D	11-87	11000	23			DRASE	532
	L11001		4		11000	23		CTR FUSELAGE	DRASE	533
I11001	Q11001		E	98	11000	23	34	29L 2531X3	DRASE	534
I11001	Q11002		E	05	11000	23	15	29L 1531X3	DRASE	535
I11001	Q11003		D		11000	21			DRASE	536
S41001	L11001	I11001	D	11-87	11000	23			DRASE	537
	L11001		4		11000			ACCESS DOORS	DRASE	538
I11001	Q11001		E	11	11000	23	17	29L 1531X0	DRASE	539
I11001	Q11002		E	63	11000	23	34	29L 2531X3	DRASE	540
I11001	Q11003		E	07	11000	23	16	29L 1531X3	DRASE	541
I11001	Q11004		D		11000	21			DRASE	542
			4		11000			AFT FUSELAGE	DRASE	543
			4		11000			F-2 11K L	DRASE	544
			4		11000			F15 MS3MA 9	DRASE	545
			4		11000			AFT FUSELAGE F15	DRASE	546
A1001	F11000	A1001	F	6	11000	21			DRASE	547
A1001	F11001	A1002	E	02	11000	21	20	29L 2431X1	DRASE	548
A1001	F11002		E	17	11000	21	18	29L 1431X1	DRASE	549
A1001	F11003		E	55	11000	21	10	29L 1531X0	DRASE	550
A1001	F11004		E	40	11000	21	16	29L 2531X3	DRASE	551
A1002	S40P	S41000	D		11000	23			DRASE	552
S41001	L11001	I11001	D	11-87	11000	23			DRASE	553
	L11001		4		11000	23		AFT FUSELAGE	DRASE	554
I11001	Q11001		E	24	11000	23	23	29L 1531X1	DRASE	555
I11001	Q11002		E	67	11000	23	25	29L 1531X3	DRASE	556
I11001	Q11003		E	05	11000	23	16	29L 1531X1	DRASE	557
I11001	Q11004		D		11000	21			DRASE	558
S41001	L11001	I11001	D	11-87	11000	23			DRASE	559
	L11001		4		11000			ACCESS DOORS	DRASE	560
I11001	Q11001		D		11000	23	19	29L 2531X3	DRASE	561
I11001	Q11002		D		11000	21			DRASE	562
			4		11000			WING ASSY F15	DRASE	563
			4		11000			CRACKS IN WINGS SKIN BEING LOOKED	DRASE	564
			4		11000			INTO BY ENGINEERS	DRASE	565
			4		11000			F-2 112 USED FOR COMPATIBILITY	DRASE	566
			4		11000			MSGT DIAGNOSE LUKE SHEET MOL PM2767	DRASE	567
			4		11000			F15 DATA SHOWS 33 MS3MA AUG75	DRASE	568
			4		11000			WING ASSY F15	DRASE	569
A1K01	F11K00	A1K01	F	21	11000	21			DRASE	570
A1K01	F11K01	A1K02	E	15	11000	21			DRASE	571
A1K01	F11K02		E	31	11000	21	120	29L 2424X0	DRASE	572
A1K01	F11K03		E	77	11000	21	29	29L 2424X0	DRASE	573
A1K01	F11K04	A1K02	E	01	11000	21	95	29L 443101	DRASE	574
A1K01	F11K05	A1K02	E	02	11000	21	25	29L 1431X1	DRASE	575
A1K01	F11K06		E	42	11000	21	10	29L 1531X0	DRASE	576
A1K01	F11K07		E	40	11000	21	18	29L 2531X3	DRASE	577
A1K02	S40P	S41K00	D		11000	23			DRASE	578
S41K01	L11K01	I11K01	D	11-85	11000	23			DRASE	579
	L11K01		4		11000			LEADING EDGE TRAILING EDGE	DRASE	580
I11K01	Q11K01		E	66	11000	23	23	29L 1531X3	DRASE	581
I11K01	Q11K02		E	34	11000	23	15	29L 2536X0	DRASE	582
I11K01	Q11K03		E		11000	21	95	29L 443101	DRASE	583
S41K01	L11K01	I11K01	D	11-85	11000	23			DRASE	584
	L11K01		4		11000			WING ACCESS PROVISIONS	DRASE	585
I11K01	Q11K01		D		11000	23	18	29L 2531X3	DRASE	586
I11K01	Q11K02		E		11000	21	28	29L 1431X1	DRASE	587
			4		11000			AIR INDUCTION SYS IS EXPECTED TO	DRASE	588
			4		11000			HAVE HIGHER FAIL RATE THAN F-2	DRASE	589
			H		11000			REASON IS MORE CYLINDERS & SERVOS	DRASE	590
			4		11000			F15 DATA SHOWS 18 MS3MA AUG75	DRASE	591

	M		11000	F4E 113 DATA USED 30 45PMA	DBASE	592
	M		11000	A COMPOSITE OF F4E & F15 FOR SMCB	DBASE	593
	4		11000	AIP INDUCTION SYS F15	DBASE	594
A1P01	F11P01	A1P01	E	22 11000 21	DBASE	595
A1P01	F11P02	A1P02	E	04 11000 21	DBASE	596
A1P01	F11P03	A1P03	E	12 11000 21	DBASE	597
A1P01	F11P04	A1P04	E	03 11000 21	DBASE	598
A1P01	F11P05	A1P05	E	07 11000 21	DBASE	599
A1P01	F11P06	A1P06	E	24 11000 21	DBASE	600
A1P01	F11P07	A1P07	E	04 11000 21	DBASE	601
A1P01	F11P08	A1P08	E	05 11000 21	DBASE	602
A1P01	F11P09	A1P09	E	12 11000 21	DBASE	603
A1P01	F11P10	A1P10	E	05 11000 21	DBASE	604
A1P12	S41P12	S41P12	D	11000 21	DBASE	605
S41P01	L11P01	I41P01	G	0293 11000 23	DBASE	606
			4	11000	DBASE	607
					VARIABLE INLET PUMP SYS	
I41P01	M11P01		D	11000*23	DBASE	608
I41P01	M11P02		D	11000 21	DBASE	609
S41P01	L11P01	I41P01	G	01655 11000 23	DBASE	610
			4	11000	DBASE	611
					CONTROLLER AIR INLET	
I41P01	M11P01		D	11000 23	DBASE	612
I41P01	M11P02		D	11000*23	DBASE	613
I41P02	M11P01	I41P01	D	11000 23	DBASE	614
I41P02	M11P02		D	07 11000 23	DBASE	615
I41P03	M11P01		D	04 11000 23	DBASE	616
I41P03	M11P02		D	34 11000 23	DBASE	617
			M	12000	DBASE	618
			4	12000	DBASE	619
			4	12000	F4E 121 DATA USED	
			4	12000	F4E 45PMA 210	
			4	12000	MSG DICELLO LUKE PH0793 SHEETMOL	
			4	12000	TSG SWEEPS LUKE 130 PH 2464	
			4	12000	COCKPIT FURNISHINGS F15	
A2001	F12001	A2001	E	02 12000 21	DBASE	621
A2001	F12002	A2002	E	03 12000 21	DBASE	622
A2001	F12003	A2003	E	07 12000 21	DBASE	623
A2001	F12004	A2004	E	07 12000 21	DBASE	624
A2001	F12005	A2005	E	07 12000 21	DBASE	625
A2001	F12006	A2006	E	07 12000 21	DBASE	626
A2001	F12007	A2007	E	07 12000 21	DBASE	627
A2002	S42P01	S42P01	D	12000 23	DBASE	628
S42001	L12001	I42001	G	00015 12000 23	DBASE	629
			4	12000	DBASE	630
					PANELS	
I42001	M12001		D	12000*23	DBASE	631
I42001	M12002		D	12000 21	DBASE	632
S42001	L12001	I42001	G	00043 12000 23	DBASE	633
			4	12000	DBASE	634
					GLAZE SHIELD INS PAN FWD CREW LH RH C	
I42001	M12001		D	12000*23	DBASE	635
I42001	M12002		D	12000 21	DBASE	636
			M	12000	DBASE	637
			4	12000	EJECTION SEAT	
			4	12000	A70 120 DATA USED	
			4	12000	ASSUME NO SEAT REMOVAL EXCEPT SCHED	
			4	12000	NO ITEM TO SHOP ITEM FAILING A70 IS	
			4	12000	NOT ON F15 F15 MS30A 122	
			4	12000	TSG JENSEN LUKE PH 2325 EXPRESS	
			4	12000	SSG SCHNEIDER EDWARDS EXPRESS	
			4	12000	MSG GREGGON LUKE PH 2325 EXPRESS VER	
			4	12000	MSG MILLER LUKE PH 2325 EXPRESS	
			4	12000	EJECTION SEAT F15	
A2001	F12001	A2001	E	125 12000 21	DBASE	640
A2001	F12002	A2002	E	12000 21	DBASE	641
A2001	F12003	A2003	E	12000 21	DBASE	642
A2003	F12001	A2001	E	66 12000 21	DBASE	643
A2003	F12002	A2002	E	16 12000 21	DBASE	644
A2003	F12003	A2003	E	17 12000 21	DBASE	645
A2002	F12001	A2001	E	12 12000 21	DBASE	646
			4	12000	DBASE	647
			4	12000	CANOPY	
			4	12000	F4E 123 DATA USED	
			4	12000	V12001 IS PRES OK	
			4	12000	HYD OPERATED CANOPY	
			4	12000	DBASE	654
			4	12000	DBASE	655
			4	12000	DBASE	656
			4	12000	DBASE	657

4	12000	HSG DICELLO LUKE PH 2753 SHEET NDL	DBASE	659						
4	12000	YSG JENSEN LUKE EXPRESS	DBASE	659						
4	12000	SSG SCHMEIDER EDWARDS EXPRESS	DBASE	660						
4	12000	RIGGING CANOPY MUCH LARGER JOB THAN	DBASE	661						
4	12000	F4 CANOPY ASSY: WILL BE DOUBLE F4E	DBASE	662						
4	12000	CANOPY MAT APPROX 2 FEET FROM 100UM	DBASE	663						
4	12000	AFTER THAT A HAND PUMP UP OF HYD MULE	DBASE	664						
4	12000	F4E MSBMA #2	DBASE	665						
4	12000	CANOPY ASSY F15	DBASE	666						
A2C13	12000	A2C13	F	25	12000	21			DBASE	667
A2C14	00447	A2C14	E	57	12000	21			DBASE	668
A2C15	00447	A2C15	E	57	12000	21			DBASE	669
A2C16	12000	A2C16	E	24	12000	21	12	29L 2421X2 1TU228	DBASE	670
A2C17	12000	A2C17	E	52	12000	21	22	29L 2431X1 1TU228	DBASE	671
A2C18	12000	A2C18	E	29	12000	21	24	29L 2422X1	DBASE	672
A2C19	12000	A2C19	E	25	12000	21	64	29L 2431X1	DBASE	673
A2C20	12000	A2C20	E	17	12000	21	19	29L 2422X1	DBASE	674
A2C21	12000	A2C21	E	23	12000	21	09	29L 2431X1	DBASE	675
A2C22	12000	A2C22	E	12	12000	21	10	29L 1531X1	DBASE	676
A2C23	12000	A2C23	E	16	12000	21	15	29L 2431X1	DBASE	677
A2C24	12000	A2C24	E	11	12000	21	50	29L 1531X3	DBASE	678
A2C25	12000	A2C25	E	65	12000	21	20	29L 2421X2 1TU228	DBASE	679
A2C26	12000	A2C26	E	35	12000	21	16	29L 2421X2 1TU228	DBASE	680
A2C27	12000	A2C27	E	25	12000	21	15	29L 2422X1 1TU228	DBASE	681
A2C28	12000	A2C28	E	23	12000	23			DBASE	682
A2C29	12000	A2C29	E	23	12000	23			DBASE	683
A2C30	12000	A2C30	E	55	12000	23	29	29L 2421X2	DBASE	684
A2C31	12000	A2C31	E	05	12000	23	17	29L 1531X3	DBASE	685
A2C32	12000	A2C32	E	06	12000	23	14	29L 2431X3	DBASE	686
A2C33	12000	A2C33	E	08	12000	23	08	29L 1530X0	DBASE	687
A2C34	12000	A2C34	E	28	12000	23	3	29L 1531X2	DBASE	688
A2C35	12000	A2C35	E	22	12000	21	22	29L 2421X2	DBASE	689
4	13400	MAIN LANDING GEAR							DBASE	690
4	13400	F4E 112 DATA USED							DBASE	692
4	13400	V TASK ARE FOR JACKING / RETRACT OPS							DBASE	693
4	13400	HYD BASICALLY F4E WITH MORE SYRUT							DBASE	694
4	13400	SPALS PROBLEMS							DBASE	695
4	13400	HY MUL BEING PROBLM HOPEFULLY							DBASE	696
4	13400	BEING CORRECTED							DBASE	697
4	13400	ADFT BEING JACKED							DBASE	698
4	13400	TEST MONTGOMERY LUKE HYD PH 2507							DBASE	699
4	13400	SSG SAMPLE EDWARDS HYD							DBASE	700
4	13400	WHEEL / TIRE 1125 ELIMINATED							DBASE	701
4	13400	MSBMA ADJUSTED ACCORDINGLY							DBASE	702
4	13400	MAIN LG F15							DBASE	703
A3A11	F13A11	A3A11	F	11	13400	21			DBASE	704
A3A12	00447	A3A12	E	04	13400	21			DBASE	705
A3A13	00447	A3A13	D		13400	21			DBASE	706
A3A14	00447	A3A14	D		13400	21			DBASE	707
A3A15	00447	A3A15	E	07	13400	21			DBASE	708
A3A16	00447	A3A16	D		13400	21			DBASE	709
A3A17	00447	A3A17	D		13400	21			DBASE	710
A3A18	00447	A3A18	D		13400	21			DBASE	711
A3A19	00447	A3A19	E	04	13400	21			DBASE	712
A3A20	00447	A3A20	E	04	13400	21			DBASE	713
A3A21	00447	A3A21	D		13400	21			DBASE	714
A3A22	00447	A3A22	D		13400	21			DBASE	715
A3A23	00447	A3A23	D		13400	21			DBASE	716
A3A24	00447	A3A24	D		13400	21	35	29L 2421X2 1060	DBASE	717
A3A25	00447	A3A25	D		13400	21	15	29L 2431X1	DBASE	718
A3A26	00447	A3A26	E	04	13400	21	26	29L 2401X2	DBASE	719
A3A27	00447	A3A27	E	09	13400	21	13	29L 1531X3	DBASE	720
A3A28	00447	A3A28	D		13400	21	47	29L 3431X1 1060	DBASE	721
A3A29	00447	A3A29	E	04	13400	21	19	29L 2431X1	DBASE	722
A3A30	00447	A3A30	D		13400	21	16	29L 2431X1 1060	DBASE	723

A3A11	M13A13	E	05	13A00	21	06	29L	1431Y1		DBASE	724		
A3A12	VJACK	13A05	0	13A00	21	17	29L	2531Y1		DBASE	725		
A3A12	VJACK	13A05	0	13A00	21	17	29L	4431Y1	1360	DBASE	726		
A3A13	VJACK	13A05	0	13A00	21					DBASE	727		
A3A14	VJACK	13A05	0	13A00	21					DBASE	728		
A3A15	VJACK	13A05	0	13A00	21	15	29L	2431Y1	1360	DBASE	729		
A3A16	VJACK	13A05	0	13A00	21					DBASE	730		
A3A17	VJACK	13A05	0	13A00	21					DBASE	731		
A3A18	SHOP	13A00	0	13A00	23					DBASE	732		
SAB101	L13A00	13A00	0	13A00	23					DBASE	732		
	L13A00	4		13A00				MAIN LG MECH COMPTS	1360	THRU	13A00	DBASE	733
IAB101	M13A00	E	51	13A00	*23	15	29L	2421Y2		DBASE	734		
IAB102	M13A00	E	25	13A00	*23	01	29L	2531Y3		DBASE	735		
IAB103	M13A00	E	25	13A00	*23	21	29L	1531Y2		DBASE	736		
IAB104	M13A00	E		13A00	21	24	29L	2421Y2		DBASE	737		
SAB101	L13A00	13A00	0	13A00	23					DBASE	739		
	L13A00	4		13A00				UPATCH MECH ETC	1360		DBASE	739	
IAB105	M13A00	E	07	13A00	*23	06	29L	1531Y2		DBASE	740		
IAB106	M13A00	E	03	13A00	*23	03	29L	2531Y3		DBASE	741		
IAB107	M13A00	E		13A00	21					DBASE	742		
		M		13A00				NOSE GEAR TO INCLUDE STEERING		DBASE	743		
		4		13A00				F4E 133 DATA USED		DBASE	744		
		4		13A00				V TASK APP FOR JACKING / RETRACT OK		DBASE	745		
		4		13A00				F15 MSBMA 149		DBASE	746		
		4		13A00				MOSE GEAR NOT THIS AOE		DBASE	747		
		M		13A00				SHIMMY PROBLEM STARTING		DBASE	748		
		4		13A00				NOSE LG / STEER F15		DBASE	749		
A3B11	F13B11	A3B11	E	25	13B00	21				DBASE	750		
A3B12	F13B12	A3B12	E	15	13B00	21				DBASE	751		
A3B13	F13B13	A3B13	E		13B00	21				DBASE	752		
A3B14	F13B14	A3B14	E		13B00	21				DBASE	753		
A3B15	F13B15	A3B15	E	05	13B00	21				DBASE	754		
A3B16	F13B16	A3B16	E		13B00	21				DBASE	755		
A3B17	F13B17	A3B17	E		13B00	21				DBASE	756		
A3B18	F13B18	A3B18	E		13B00	21				DBASE	757		
A3B19	F13B19	A3B19	E	04	13B00	21				DBASE	758		
A3B20	F13B20	A3B20	E		13B00	21				DBASE	759		
A3B21	F13B21	A3B21	E		13B00	21				DBASE	760		
A3B22	F13B22	A3B22	E		13B00	21				DBASE	761		
A3B23	F13B23	A3B23	E		13B00	21				DBASE	762		
A3B24	F13B24	A3B24	E		13B00	21				DBASE	763		
A3B25	F13B25	A3B25	E		13B00	21	39	29L	2421Y2	1360	1TU028	DBASE	764
A3B26	F13B26	A3B26	E		13B00	21	20	29L	2423Y3	1360	1TU228	DBASE	765
A3B27	F13B27	A3B27	E		13B00	21	11	29L	1431Y1	1360	1TU228	DBASE	766
A3B28	F13B28	A3B28	E	31	13B00	21	13	29L	2421Y2		DBASE	767	
A3B29	F13B29	A3B29	E	02	13B00	21	14	29L	1531Y3		DBASE	768	
A3B30	F13B30	A3B30	E	02	13B00	21	05	29L	1431Y1		DBASE	769	
A3B31	F13B31	A3B31	E	03	13B00	21	13	29L	2531Y3		DBASE	770	
A3B32	F13B32	A3B32	E		13B00	21	10	29L	2431Y1		DBASE	771	
A3B33	F13B33	A3B33	E	31	13B00	21	33	29L	2423Y3		DBASE	772	
A3B34	VJACK	13B00	0	13B00	21					DBASE	773		
A3B35	VJACK	13B00	0	13B00	21					DBASE	774		
A3B36	VJACK	13B00	0	13B00	21					DBASE	775		
A3B37	VJACK	13B00	0	13B00	21					DBASE	776		
A3B38	VJACK	13B00	0	13B00	21	15	29L	2431Y1		DBASE	777		
A3B39	VJACK	13B00	0	13B00	21					DBASE	778		
A3B40	VJACK	13B00	0	13B00	21					DBASE	779		
SAB101	L13B00	13B00	0	13B00	21					DBASE	780		
	L13B00	4		13B00				NOSE GEAR / STEP COMPTS		DBASE	780		
IAB300	M13B00	E	03	13B00	*23	00	29L	2421Y2		DBASE	781		
IAB301	M13B00	E	76	13B00	*23	10	29L	1421Y2		DBASE	782		
IAB302	M13B00	E	07	13B00	*23	10	29L	1531Y3		DBASE	783		
IAB303	M13B00	E	12	13B00	*23	05	29L	2531Y3		DBASE	784		
IAB304	M13B00	E		13B00	21	39	29L	2423Y3		DBASE	785		
		M		13C00				ARRESTING HOOK SYS		DBASE	786		
		4		13C00				F4E 135YX DATA USED		DBASE	787		
		M		13C00				F15MSBMA NO FAILURE		DBASE	788		
		M		13C00				ARREST HOOK SYS F15		DBASE	789		

A3001	F13001	A3001	F	16	13000	21				DBASE	791	
A3001	F13001	A3002	F	4	13000	21	33	29L	2-21X2	DBASE	791	
A3001	F13001		F	22	13000	21	16	29L	2-21X2	DBASE	792	
A3001	F13001		F	07	13000	21	09	29L	2-23X0	DBASE	793	
A3001	F13001	A3002	F	04	13000	21	22	29L	1-31X1	DBASE	794	
A3001	F13002		F	05	13000	21	13	29L	1-31X1	DBASE	795	
A3001	F13003		F	21	13000	21	19	29L	1531X0	DBASE	796	
A3001	F13004		F	02	13000	21	16	29L	2631X3	DBASE	797	
A3002	SHOP	SA3003	D		13000	23				DBASE	798	
SA3001	L13001	IA3001	G	156	13000	23				DBASE	799	
IA3001	F13001		H		13000				ARRESTING HOOK ASY	DBASE	800	
IA3001	F13001		F	5	13000	*23	03	29L	2-21X2	DBASE	801	
IA3001	F13001		F	31	13000	*23	15	29L	1-21X2	DBASE	802	
IA3001	F13001		F	15	13000	*23	13	29L	1531X0	DBASE	803	
IA3001	F13001		F		13000	21	33	29L	2-21X2	DBASE	804	
SA3001	L13001	IA3001	G	156	13000	23				DBASE	805	
IA3001	F13001		H		13000				DAMPER ASY	DBASE	806	
IA3001	F13001		F	32	13000	*23	03	29L	1531X0	DBASE	807	
IA3001	F13001		F	61	13000	*23	21	29L	1531X1	DBASE	808	
IA3001	F13001		F		13000	21	32	29L	1-31X1	DBASE	809	
SA3001	L13001	IA3001	G	127	13000	23			NDC	DBASE	810	
IA3001	F13001		D		13000	*23	20	29L	1531X3	DBASE	811	
IA3001	F13001		D		13000	21				DBASE	812	
			H		13000				BRAKE SYS	DBASE	813	
			H		13000				F4E 1341 1344 DATA USED	DBASE	814	
			H		13000				INCLUDES JACKING AS NECESSARY	DBASE	815	
			H		13000				F15 MSRMA 160	DBASE	816	
			H		13000				BRAKE SYS F15	DBASE	817	
A3001	F13001	A3001	F	65	13000	21				DBASE	818	
A3001	000007	A3001	E	63	13000	21				DBASE	819	
A3001	000007	A3001	D		13000	21				DBASE	820	
A3001	000007	A3001	D		13000	21				DBASE	821	
A3001	000007	A3001	D		13000	21				DBASE	822	
A3001	000007	A3001	E	24	13000	21				DBASE	823	
A3001	000007	A3001	D		13000	21				DBASE	824	
A3001	000007	A3001	D		13000	21				DBASE	825	
A3001	000007	A3001	D		13000	21				DBASE	826	
A3001	000007	A3001	D		13000	21				DBASE	827	
A3001	F13001		E	06	13000	21	14	29L	1-31X1	DBASE	828	
A3001	F13001		E	07	13000	21	10	29L	1531X0	DBASE	829	
A3001	F13001	A3001	D		13000	21	16	29L	2-31X1 1060	1TU228	DBASE	830
A3001	VJACK		D		13000	21				DBASE	831	
A3001	F13001		E	16	13000	21	25	29L	2-21X2 1060	1TU228	DBASE	832
A3001	F13001	A3001	E	6	13000	21	40	29L	2-21X2 1060	1TU228	DBASE	833
A3001	VJACK	A3001	D		13000	21				DBASE	834	
A3001	SHOP	SA3001	D		13000	21				DBASE	835	
SA3001	L13001	IA3001	D		13000	21				DBASE	836	
IA3001	F13001		H		13000				BRAKE COMPTS	DBASE	837	
IA3001	F13001		F	5	13000	*23	10	29L	1531X1	DBASE	838	
IA3001	F13001		F	16	13000	*23	02	29L	1531X1	DBASE	839	
IA3001	F13001		F	77	13000	*23	51	29L	1-31X1 1531X1	DBASE	840	
IA3001	F13001		D		13000	21				DBASE	841	
			H		13000				LANDING CNTL / WARNING / EMERG SYS	DBASE	842	
			H		13000				F4E 1341 DATA USED FOR F15 137 / 10F	DBASE	843	
			H		13000				F15 MSRMA 160	DBASE	844	
			H		13000				LNG CNTL WARN / EMERG SYS F15	DBASE	845	
A3001	F13001	A3001	F	178	13000	21				DBASE	846	
A3001	000007	A3001	D		13000	21				DBASE	847	
A3001	CALGRP	A3001	D		13000	21				DBASE	848	
A3001	F13001	A3001	E	24	13000	21	23	29L	2-21X2 1060	DBASE	849	
A3001	F13001		E	02	13000	21	12	29L	1531X0	DBASE	850	
A3001	F13001		F	07	13000	21	08	29L	2-21X2 1060	DBASE	851	
A3001	F13001	A3001	E	14	13000	21	22	29L	2-23X0 1060	DBASE	852	
A3001	F13001		E	27	13000	21	16	29L	1-23X0 1060	DBASE	853	
A3001	F13001	A3001	E	10	13000	21	10	29L	4-31X1	DBASE	854	
A3001	F13001		D		13000	21	40	29L	2-21X2 1060	DBASE	855	

A3F00	R13F01	E	23	13F00	21	07 29L 2326R2 1060	DBASE	856	
A3F02	V13F01	J		13F00	21	11 29L 2-21X2 1060 1421X1	DBASE	857	
A3F05	V13F02	E	1.	13F00	21	05 29L 1-31X1 1060	DBASE	858	
A3F07	S40F	C13F01	J	13F00	23		DBASE	859	
S43F01	L13F01	I13F01	J	13F00	23		DBASE	860	
I13F01	413F01	J	4	13F00	23	110 CONT M474 / F4500 SYS	DBASE	861	
I13F01	413F01	J	4	13F00	23	12 29L 1-21X2	DBASE	862	
I13F01	413F01	J	4	13F00	23		DBASE	863	
		M		13M00		SKID CONT SYS	DBASE	864	
		M		13M00		F15 DATA USED	DBASE	865	
		M		13M00		F15 MS14A 03 AUG 75	DBASE	866	
		M		13M00		SKID CONT SYS F15	DBASE	867	
A3M01	R13M01	A3M01	F	237	13M00	21	DBASE	868	
A3M02	V13M01	A3M01	J		13M00	21	31 29L 2-23X0 1-31X1 1-21X2	DBASE	869
A3M03	R13M02	E	02	13M00	21	34 29L 2-21X2	DBASE	870	
A3M04	R13M03	E	01	13M00	21	11 29L 2-21X2	DBASE	871	
A3M05	R13M04	A3M02	E	35	13M00	21	14 29L 2-23X0	DBASE	872
A3M06	R13M05	E	55	13M00	21	12 29L 2-23X0	DBASE	873	
A3M07	S40F	S13M01	J		13M00	23	DBASE	874	
S13M01	L13M01	I13M01	J		13M00	23	DBASE	875	
L13M01	413M01	J	4	13M00		CONT SYS	DBASE	876	
I13M01	413M01	J	4	13M00	23	11 29L 1-23X0	DBASE	877	
I13M01	413M01	J	4	13M00	21	14 29L 2-23X0	DBASE	878	
		M		13J00		MHEEL / TIRE	DBASE	879	
		M		13J00		F15 1325 1325 1326 FOR F15 13AJ 01	DBASE	880	
		M		13J00		ASSUME F15 1325/1326 CHGS EQUAL F15	DBASE	881	
		M		13J00		WARTIME CHGS	DBASE	882	
		M		13J00		F15 MS14A 15	DBASE	883	
		M		13J00		MHEEL / TIRE F15	DBASE	884	
A3J01	R13J01	A3J01	F	19	13J00	21	DBASE	885	
A3J02	V13J01	A3J02	J		13J00	21	09 29L 2-31X1	DBASE	886
A3J03	S40F	S13J01	J		13J00	23	DBASE	887	
S13J01	L13J01	I13J01	J		13J00	23	DBASE	888	
L13J01	413J01	J	4	13J00		MHEEL / TIRE	DBASE	889	
I13J01	413J01	J	4	13J00	23	02 29L 1-401Y1	DBASE	890	
I13J01	413J01	J	4	13J00	23	12 29L 1-71Y1	DBASE	891	
I13J01	413J01	J	4	13J00	23	20 29L 2-401Y1	DBASE	892	
I13J01	413J01	J	4	13J00	23	15 29L 2-71Y1	DBASE	893	
		M		14A00		CONTROL STICK	DBASE	894	
		M		14A00		F15 MS14A 21	DBASE	895	
		M		14A00		SSG THOMPSON FLT CONT EDWARDS	DBASE	896	
		M		14A00		TSG PAUL LUKE FLT CONT	DBASE	897	
		M		14A00		TSG FULTON EDWARDS	DBASE	898	
		M		14A00		TSG MCWILLON LUKE HYD	DBASE	899	
		M		14A00		SSG SAMPLE EDWARDS HYD	DBASE	900	
		M		14A00		MSG THEED EDWARDS APG	DBASE	901	
		M		14A00		TSG SWEEBE LUKE APG 2484	DBASE	902	
		M		14A00		F111F 1325 DATA USED	DBASE	903	
		M		14A00		CONTROL STICK F15	DBASE	904	
A4A01	R14A01	A4A01	F	97	14A00	21	DBASE	905	
A4A02	V14A01	A4A02	J		14A00	21	DBASE	906	
A4A03	R14A02	A4A03	J		14A00	21	DBASE	907	
A4A04	R14A03	A4A04	J		14A00	21	DBASE	908	
A4A05	R14A04	A4A05	J		14A00	21	DBASE	909	
A4A06	R14A05	A4A06	J		14A00	21	23 29L 2-23X0 1060 1TU228	DBASE	910
A4A07	R14A06	A4A07	J	55	14A00	21	06 29L 2326R2	DBASE	911
A4A08	R14A07	A4A08	J	45	14A00	21	15 29L 2-23X0 1060 1TU228	DBASE	912
A4A09	R14A08	A4A09	J		14A00	21	18 29L 2326R2 1060 1TU228	DBASE	913
A4A10	S40F	S14A01	J		14A00	23	DBASE	914	
S14A01	L14A01	I14A01	J		14A00	23	DBASE	915	
L14A01	414A01	J	4	14A00		CONTROL STICK ASSY	DBASE	916	
I14A01	414A01	J	4	14A00	23	04 29L 1-23X0	DBASE	917	
I14A01	414A01	J	4	14A00	23	11 29L 1326R1	DBASE	918	
I14A01	414A01	J	4	14A00	23	20 G	DBASE	919	
I14A01	414A01	J	4	14A00	21	15 29L 2-23X0	DBASE	920	
		M		14A00		F15 DATA USED	DBASE	921	

	M		14430	NO COMPARIBILITY				09ASE	922
	M		14431	TSGT FOGLE LUKE				09ASE	923
	M		14432	TSGT FULTON EDWARDS				09ASE	924
	M		14433	F15 MSPMA 420				09ASE	925
	M		14434	PITCH & ROLL CHANNEL ASY F15				09ASE	926
A4431	F1-431	1-431	F	711	14431	21		09ASE	927
A4431	R1-431	1-431	D		14431	21	54 29L 2421X2	09ASE	928
A4432	V1-431	1-431	D		14431	21	40 29L 1-21X2 232632 247:01 1060	09ASE	929
A4433	S40P	SA4433	D		14433	23		09ASE	930
SA4431	L1-431	1-431	D		14431	23		09ASE	931
	L1-431		M		14430		PITCH & ROLL CHANNEL ASY	09ASE	932
IA4431	N1-431		D		14431	23	19 29L 1-21X2	09ASE	933
IA4431	Q1-431		I		14431	21	54 29L 2-21X2	09ASE	934
			M		14000		STAR SYS DATA	09ASE	935
			M		14000		F111F 1400A + 37% 1-431 + 14000 C 30	09ASE	936
			M		14000		F4E 1-311 FOR APG & 11% OF SHEET MOL	09ASE	937
			M		14000		F15 MSPMA 420	09ASE	938
			M		14000		STAR SYS F15	09ASE	939
A4001	F1-001	1-001	F	150	14001	21		09ASE	940
A4001	D3R4G7	1-001	D		14001	21		09ASE	941
A4002	CA LGPU	1-002	D		14002	21		09ASE	942
A4003	D3R4H7	1-003	D		14003	21		09ASE	943
A4004	CA LTTJ	1-004	D		14004	21		09ASE	944
A4005	R1-001	1-001	E	50	14001	21	26 29L 2-31X1 1060 170224	09ASE	945
A4006	R1-002	1-002	E	24	14002	21	17 29L 1431X1 1060 170225	09ASE	946
A4007	T1-001	1-001	E	05	14001	21	20 29L 232632 1060 170224	09ASE	947
A4008	R1-003	1-003	E	1	14003	21	57 29L 2421X2 170226	09ASE	948
A4009	S40P	SA4009	D		14009	23		09ASE	949
SA4001	L1-001	1-001	D		14001	23		09ASE	950
	L1-001		M		14000		STAR SYS	09ASE	951
IA4001	N1-001		D		14001	23	15 29L 2421X2	09ASE	952
IA4001	Q1-001		I		14001	21	26 29L 2431X1	09ASE	953
			M		14000		RUDDER	09ASE	954
			M		14000		F111F 14400 23X F4E 1-42 C 0 14400	09ASE	955
			M		14000		FOR APG	09ASE	956
			M		14000		F15 MSPMA 32	09ASE	957
			M		14000		RUDDER F15	09ASE	958
A4001	F1-001	1-001	F	21	14001	21		09ASE	959
A4001	D3R4G7	1-001	D		14001	21		09ASE	960
A4002	D3R4H7	1-002	D		14002	21		09ASE	961
A4003	CA LGPU	1-003	D		14003	21		09ASE	962
A4004	CA LTTJ	1-004	D		14004	21		09ASE	963
A4005	R1-001	1-001	E	20	14001	21	35 29L 2431X1 1060 170224	09ASE	964
A4006	R1-002	1-002	E	61	14002	21	30 29L 2421X2 1060 170224	09ASE	965
A4007	T1-001	1-001	E	06	14001	21	12 29L 1431X1	09ASE	966
A4008	T1-002	1-002	E	05	14001	21	25 29L 232632 1060 170225	09ASE	967
A4009	R1-002	1-002	E	02	14002	21	20 29L 2431X1	09ASE	968
A4010	S40P	SA4010	D		14010	23		09ASE	969
SA4001	L1-001	1-001	D		14001	23		09ASE	970
	L1-001		M		14000		RUDDER COMPTS	09ASE	971
IA4001	N1-001		E	97	14001	23	10 29L 2421X2	09ASE	972
IA4001	W1-001		E	03	14001	23	16 29L 2421X2	09ASE	973
IA4001	Q1-001		I		14001	21	77 29L 2421X2	09ASE	974
			M		14000		SPEED BRAKE SYS	09ASE	975
			M		14000		F4E 14000 LEAS 14000 DATA 0000	09ASE	976
			M		14000		F15 MSPMA 210	09ASE	977
			M		14000		SPEED BRAKE SYS F15	09ASE	978
A4E01	F1-001	1-001	F	67	14E01	21		09ASE	979
A4E01	D3R4G7	1-001	D		14E01	21		09ASE	980
A4E02	CA LGPU	1-002	D		14E02	21		09ASE	981
A4E03	D3R4H7	1-003	D		14E03	21		09ASE	982
A4E04	CA LTTJ	1-004	D		14E04	21		09ASE	983
A4E05	R1-001	1-001	E	06	14E01	21	40 29L 2421X2 1060 170228	09ASE	984
A4E06	R1-002	1-002	E	56	14E02	21	16 29L 2421X2 1060 170228	09ASE	985
A4E07	T1-001	1-001	E	02	14E01	21	17 29L 2421X2 1060 170228	09ASE	986
A4E08	R1-002	1-002	E	05	14E02	21	11 29L 2423X2 1060 170228	09ASE	987

A4E10	21-212	A4E02	E	09	14E00	21	30	29L	2431X1	1060	1TU228	DBASE	988
A4E10	41-212		F	07	14E10	21	12	29L	1531X1	1060	1TU228	DBASE	989
A4E10	41-213		E	03	14E10	21	10	29L	1531X0	1060	1TU228	DBASE	990
A4E10	41-214		F	02	14E00	21	21	29L	2531X3	1060	1TU228	DBASE	991
A4E12	V1-211		D		14E10	21	10	29L	2421X2	1060	1TU228	DBASE	992
A4E12	SHDP	S4-211	D		14E10	21						DBASE	993
S4E11	L14E11	S4-211	D	0215	14E00	23						DBASE	994
	L14E11		F		14E10	21						DBASE	995
I4E11	41-211		E	50	14E00	23	50	29L	2531X3			DBASE	996
I4E11	41-212		E	47	14E10	23	44	29L	1531X3			DBASE	997
I4E11	41-213		D		14E00	21						DBASE	998
S4E11	L14E11	I4-211	D	0211	14E00	23						DBASE	999
	L14E11		F		14E10	21						DBASE	1000
I4E11	41-214		E	50	14E00	23	17	29L	2421X2			DBASE	1001
I4E11	41-215		E	50	14E00	23	17	29L	1421X2			DBASE	1002
I4E11	41-216		D		14E00	21						DBASE	1003
			F		14G10				LATERAL SYS			DBASE	1004
			F		14G11				FLAP 14-228 ACTUATOR F4E 14-228 LAT HY			DBASE	1005
			F		14G01				F15 MSPMA +2L			DBASE	1006
			F		14G01				LATERAL SYS F15			DBASE	1007
A4G11	F14G11	A4G11	F	533	14G10	21						DBASE	1008
A4G11	22-217	A-211	D		14G01	21						DBASE	1009
A4G11	20-217	A-211	D		14G01	21						DBASE	1010
A4G11	21-217	A-211	D		14G01	21						DBASE	1011
A4G11	21-217	A-211	D		14G01	21						DBASE	1012
A4G12	41-211	A-211	E	41	14G11	21	36	29L	2421X2	1060	1TU228	DBASE	1013
A4G12	41-212	A-211	D		14G11	21	6	29L	2421X2	1060	1TU228	DBASE	1014
A4G12	41-213	A-211	E	59	14G11	21	21	29L	2421X2			DBASE	1015
A4G13	3-21P	S4-211	D		14G00	23						DBASE	1016
S4G11	L14G11	I4-211	D		14G00	23						DBASE	1017
	L14G11		F		14G10				LATERAL SYS COMPTS			DBASE	1018
I4-G11	41-211		E	94	14G01	23	15	29L	2421X2			DBASE	1019
I4-G11	41-212		E	06	14G01	23	17	29L	2421X2			DBASE	1020
I4-G11	01-211		D		14G01	21						DBASE	1021
			F		14H10				FLAP SYS			DBASE	1022
			F		14H00				F4E 14-228 DATA USED 14E2N NOTHING			DBASE	1023
			F		14H10				F15 MSPMA NO FAILURE			DBASE	1024
			F		14H10				FLAP SYS F15			DBASE	1025
A4H11	F14H11	A4H11	F	+26	14H10	21						DBASE	1026
A4H11	22-217	A-211	D		14H10	21						DBASE	1027
A4H11	20-217	A-211	D		14H10	21						DBASE	1028
A4H11	21-217	A-211	D		14H10	21						DBASE	1029
A4H11	21-217	A-211	D		14H10	21						DBASE	1030
A4H12	41-211	A-211	E	24	14H11	21	30	29L	2421X2	1060	1TU228	DBASE	1031
A4H12	41-212	A-211	E	76	14H11	21	21	29L	2421X2	1060	1TU228	DBASE	1032
A4H12	41-213	A-211	D		14H11	23						DBASE	1033
S4H11	L14H11	I4-211	D		14H00	23						DBASE	1034
	L14H11		F		14H10				FLAP SYS COMPTS			DBASE	1035
I4H11	41-211		E	50	14H00	23	10	29L	1421X2			DBASE	1036
I4H11	41-212		E	40	14H00	23	40	29L	2421X2			DBASE	1037
I4H11	41-213		E	10	14H00	23	24	29L	1536X0			DBASE	1038
I4H11	01-211		D		14H00	21						DBASE	1039
			F		23000				THIS IS F-105 ENGINE NETWORK AS			DBASE	1040
			F		23000				DEFINED BY SGT HODGES, 10T WAPMATH. 14-228			DBASE	1041
			F		23000				AND MR ROFF AT LUNE 24-21			DBASE	1042
			F		23000				CLOCK FOR FIRST ENGINE			DBASE	1043
93000	F23000	93001	F	5	23000	21						DBASE	1044
	F27000		F		27000				CLOCK FOR SECOND ENGINE			DBASE	1045
87000	F27000	93001	F	5	27000	21						DBASE	1046
			F		23000				V23ARM MUNITIONS SAFE			DBASE	1047
03001	V23ARM	93002	E	11	23000	21	3	29L	1450X0			DBASE	1048
			F		23000				T23000 TO NOTIFY PROBLEM			DBASE	1049
03001	T23000	93002	E	05	23000	21	5	29L	2430X0 1421X1			DBASE	1050
			F		23000				023104 TOW AHEAD TO -A0			DBASE	1051
93002	O23104	93004	D		23000	21	5	29L	1431X1			DBASE	1052
			F		23000				O23PAN REM OR REPLACE ENGINE PANELS			DBASE	1053

83204	02324M	83206	D	23100	21	3 29L 1431X1	0BASE	1054
			4	23100		02324Q INSTALL OR REMOVE AOPT FROM PAC	0BASE	1055
83205	02324D	83206	J	23100	21	3 29L 1432X0 1432X1	0BASE	1056
			4	23100		T23001 TROUBLESHOOT, FIM, TRIM, AND LEAK	0BASE	1057
			M	23100		NO PARTS REPLACEMENT	0BASE	1058
83206	T23001	83207	E	25	23100	30 29L 1432X0 1431X1	0BASE	1059
			4	23100		M23000 HINDR REPAIR WITH PART CHANGES	0BASE	1060
			4	23100		ON TRIM PAD	0BASE	1061
83208	M23000	83206	E	50	23100	50 29L 1432X0 1431X1	0BASE	1062
83209	M23000	83206	F	25	23100		0BASE	1063
83210	02324D	83206	J	23100	21		0BASE	1064
83212	02324M	83206	J	23100	21		0BASE	1065
83214	02320M		O	23100	21		0BASE	1066
			4	23100		02324K ENGINE LEAK CHECK	0BASE	1067
83215	02324K	83206	J	23100	21	5 29L 2432X0 1431X1	0BASE	1068
83216	02324D	83206	J	23100	21		0BASE	1069
83218	02324D	83206	J	23100	21		0BASE	1070
83219	02320M	83206	J	23100	21		0BASE	1071
			4	23100		M23001 HINDR REPAIR WITH PART CHANGE	0BASE	1072
			4	23100		ON FLIGHT LINE	0BASE	1073
83222	M23001	83206	E	90	23100	10 29L 2432X0	0BASE	1074
83223	M23001	83206	E	10	23100		0BASE	1075
83224	02324K		O	23100	21		0BASE	1076
83225	02324K		O	23100	21		0BASE	1077
83231	02324M	83206	J	23100	21		0BASE	1078
83232	02324M	83206	J	23100	21		0BASE	1079
83234	M23001		I	23100	21	15 29L 4432X0	0BASE	1080
83235	02324K	83206	J	23100	21	15 29L 4432X0	0BASE	1081
83236	T23001	83206	J	23100	21		0BASE	1082
83238	02324M		E	95	23100		0BASE	1083
83239	02320M	83206	E	65	23100		0BASE	1084
			4	23100		ENGINE IN SHOP AND TEST CELL	0BASE	1085
83241	02324D	83206	J	23100	23	35 29L 3432X0 19300P	0BASE	1086
			4	23100		DTOW23 TOW ENGINE TO TEST CELL	0BASE	1087
83242	DTOW23	83206	E	20	23100	1 29L 2432T3	0BASE	1088
83243	SHOP	83206	E	75	23100		0BASE	1089
			4	23100		02310L CONNECT AND DISCONNECT TEST CELL	0BASE	1090
83244	02320L	83206	J	23100	23	5 29L 5432T3	0BASE	1091
			4	23100		T23002 TROUBLESHOOT AND TRIM TEST CELL	0BASE	1092
83245	T23002	83206	J	23100	23	50 29L 4432T3	0BASE	1093
			4	23100		T23003 TEND AND SOAP TEST CELL	0BASE	1094
83246	T23003	83206	E	65	23100	5 29L 4432T3	0BASE	1095
83248	02320L	83206	E	35	23100		0BASE	1096
83249	02320L	83206	J	23100	23		0BASE	1097
83250	SHOP	83206	J	23100	23		0BASE	1098
83251	DTOW23	83206	J	23100	23		0BASE	1099
			4	23100		M23002 FINAL SHOP WORK GOOD ENGINE OUT	0BASE	1100
83252	M23002		J	23100	23	40 29L 4432X0	0BASE	1101
83253	DTOW23	83206	J	23100	23		0BASE	1102
			4	23100		M23001 ENGINE SHOP WORK	0BASE	1103
83254	M23001	83206	J	23100	23	50 29L 4432X0	0BASE	1104
			4	23100		M23002 ENGINE SHOP WORK	0BASE	1105
83255	M23002	83206	E	90	23100	60 29L 5432X0	0BASE	1106
			4	23100		M23003 ENGINE SHOP WORK	0BASE	1107
83256	M23003	83206	E	10	23100	320 29L 4432X0	0BASE	1108
	L23A10		4	23100	23	INLET FAN MODULE	0BASE	1109
83257	L23A10	83206	E	10516	23100		0BASE	1110
83258	M23A10		E	27	23100	40 29L 3432Y0	0BASE	1111
83259	M23A10		E	2	23100	22 29L 1531X1	0BASE	1112
83260	M23A10		E	55	23100	5 29L 2432X0	0BASE	1113
83261	M23A10		E	84	23100	5 29L 2432T3	0BASE	1114
83262	02324D		O	23100	21		0BASE	1115
	L23A10		4	23100	23	CURE ENGINE MODULE	0BASE	1116
83263	L23A10	83206	E	33967	23100		0BASE	1117
83264	M23A10		E	12	23100	13 29L 1432X0	0BASE	1118
83265	M23A10		E	15	23100	16 29L 1432X0	0BASE	1119

8300	N230	E	73	23000*23	5	29L	1432X0	DBASE	1123
8301	Q230	E		23000 21				DBASE	1124
	L230	4		23000 23			FAN-DRIVE TURBINE MODULE	DBASE	1125
S9300	L230	37000	3	23000 21				DBASE	1126
9300	M230	E	46	23000*23	90	29L	1432X0	DBASE	1127
9301	K230	E	1	23000*23	3	29L	1432X0	DBASE	1128
9302	N230	E	51	23000*23	10	29L	1432X0	DBASE	1129
9303	Q230	E		23000 21				DBASE	1130
	L230	4					AUGMENTOR DUST AND NOZZLE MODULE	DBASE	1131
S9300	L230	35000	6	23000 23				DBASE	1132
9300	M230	E	13	23000*23	10	29L	1432X0	DBASE	1133
9301	K230	E	11	23000*23	7	29L	1432X0	DBASE	1134
9302	N230	E	4	23000*23	45	29L	1432X0	DBASE	1135
9303	Q230	E	72	23000*23	5	29L	1432X0	DBASE	1136
	L230	4		23000 21				DBASE	1137
							GEARBOX MODULE	DBASE	1138
S9300	L230	33000	5	23000 23				DBASE	1139
9300	M230	E	11	23000*23	5	29L	1432X0	DBASE	1140
9301	K230	E	56	23000*23	17	29L	1432X0	DBASE	1141
9302	N230	E		23000 21				DBASE	1142
	L230	4					FUEL SYSTEM	DBASE	1143
S9300	L230	33000	5	23000 23				DBASE	1144
9300	M230	E		23000*23				DBASE	1145
9301	K230	E		23000*23				DBASE	1146
9302	N230	E		23000*23				DBASE	1147
9303	Q230	E		23000 21				DBASE	1148
	L230	4					OIL SYSTEM	DBASE	1149
S9300	L230	33000	5	23000 23				DBASE	1150
9300	M230	E		23000*23				DBASE	1151
9301	K230	E		23000*23				DBASE	1152
9302	N230	E		23000*23				DBASE	1153
9303	Q230	E		23000 21				DBASE	1154
	L230	4					ELECTRICAL SYSTEM	DBASE	1155
S9300	L230	33000	5	23000 23				DBASE	1156
9300	M230	E		23000*23				DBASE	1157
9301	K230	E		23000*23				DBASE	1158
9302	N230	E		23000*23				DBASE	1159
9303	Q230	E		23000 21				DBASE	1160
	L230	4					AUGMENTOR NOZZLE ACTUATION SYSTEM	DBASE	1161
S9300	L230	33000	5	23000 23				DBASE	1162
9300	M230	E		23000*23				DBASE	1163
9301	K230	E		23000*23				DBASE	1164
9302	N230	E		23000*23				DBASE	1165
9303	Q230	E		23000 21				DBASE	1166
	L230	4					MISC COMPONENTS	DBASE	1167
S9300	L230	33000	5	23000 23				DBASE	1168
9300	M230	E		23000*23				DBASE	1169
9301	K230	E		23000*23				DBASE	1170
9302	N230	E		23000*23				DBASE	1171
9303	Q230	E		23000 21				DBASE	1172
	L230	4					ENGINE INSTRUMENTS	DBASE	1173
9300	M230	E		23000*23				DBASE	1174
9301	K230	E		23000*23				DBASE	1175
9302	N230	E		23000*23				DBASE	1176
9303	Q230	E		23000 21				DBASE	1177
	L230	4					JET FUEL STARTER	DBASE	1178
				24000			F15 INFO AND INTERVIEW USED	DBASE	1179
				24000			Y24000 SHOP REPAIR	DBASE	1180
				24000			Y24000 TEST CELL RUN	DBASE	1181
				24000			TSGT PETERSON ENG PH 2744	DBASE	1182
				24000			TSGT GOODMAN ENG PH 2974	DBASE	1183
				24000			JET FUEL STARTER F15	DBASE	1184
8400	F2400	F	44	24000 21				DBASE	1185

8441	0034H7	8441A	J	2400	21				DBASE 1186
8441	0034H7	8441B	J	2400	21				DBASE 1187
8441	0034H7	8441C	J	2400	21	25	29L	1-23X0 1-32X0 1-31X1	DBASE 1188
8441	0034H7	8441D	J	2400	21	3	29L	1-31X1	DBASE 1189
8441	0034H7	8441E	A	2400	21	4	29L	1-31X1	DBASE 1190
8441	0034H7	8441F	E	2400	21	33	29L	2-32X0 1TU225	DBASE 1191
8441	0034H7	8441G	E	22	2400	21	17	29L 2-21X2 1TU225	DBASE 1192
8441	0034H7	8441H	E	2	2400	21	35	29L 2-32X1 1TU225	DBASE 1193
8441	0034H7	8441I	J	2400	21	24	29L	1-32X0	DBASE 1194
8441	0034H7	8441J	J	2400	21	33	29L	2-32X0	DBASE 1195
8441	0034H7	8441K	J	2400	21	21	29L	2-32X0	DBASE 1196
8441	0034H7	8441L	J	2400	23				DBASE 1197
8441	0034H7	8441M	J	2400	23				DBASE 1198
8441	0034H7	8441N	J	2400	23			JFS SYS	DBASE 1199
8441	0034H7	8441O	J	2400	23	15	29L	1-32X0	DBASE 1200
8441	0034H7	8441P	J	2400	23			ACC DR GEARBOXES	DBASE 1201
8441	0034H7	8441Q	J	2400	23			F15 DATA USED	DBASE 1202
8441	0034H7	8441R	J	2400	23			NO WAY TO TEST 4M4DS	DBASE 1203
8441	0034H7	8441S	J	2400	23			NEW PTO SHAFT SEAL IS WORKING	DBASE 1204
8441	0034H7	8441T	J	2400	23			MSBMA ACC FROM 25 TO 75 FOR CORRECT-	DBASE 1205
8441	0034H7	8441U	J	2400	23			TIVE ACTIONS	DBASE 1206
8441	0034H7	8441V	J	2400	23			ACC DR GEARBOX F15	DBASE 1207
8441	0034H7	8441W	J	25	2400	21			DBASE 1208
8441	0034H7	8441X	J	2400	21				DBASE 1209
8441	0034H7	8441Y	J	2400	21	5	29L	1-31X1	DBASE 1210
8441	0034H7	8441Z	J	2400	21				DBASE 1211
8441	0034H7	8441AA	J	2400	21	18	29L	2-31X1	DBASE 1212
8441	0034H7	8441AB	J	21	2400	21	24	29L 2-32X0 1TU225	DBASE 1213
8441	0034H7	8441AC	J	5	2400	21	10	29L 1-31X1	DBASE 1214
8441	0034H7	8441AD	J	53	2400	21	23	29L 2-32X0 1TU225	DBASE 1215
8441	0034H7	8441AE	J	2400	23				DBASE 1216
8441	0034H7	8441AF	J	2400	23				DBASE 1217
8441	0034H7	8441AG	J	2400	23			ACC DR GEARBOX	DBASE 1218
8441	0034H7	8441AH	J	2400	23	11	29L	1-32X0	DBASE 1219
8441	0034H7	8441AI	J	2400	21	23	29L	2-32X1	DBASE 1220
8441	0034H7	8441AJ	J	2400	21			JFS STARTING SYS	DBASE 1221
8441	0034H7	8441AK	J	2400	21			NO COMPATIBILITY AVAILABLE	DBASE 1222
8441	0034H7	8441AL	J	2400	21			USED F15 DATA	DBASE 1223
8441	0034H7	8441AM	J	2400	21			JFS START SYS F15	DBASE 1224
8441	0034H7	8441AN	J	355	2400	21			DBASE 1225
8441	0034H7	8441AO	J	2400	21	14	29L	2-21X2 1TU225 1-31X1	DBASE 1226
8441	0034H7	8441AP	J	2400	23				DBASE 1227
8441	0034H7	8441AQ	J	2400	23				DBASE 1228
8441	0034H7	8441AR	J	2400	23			JFS START SYS	DBASE 1229
8441	0034H7	8441AS	J	2400	23	22	29L	2-21X2	DBASE 1230
8441	0034H7	8441AT	J	2400	21				DBASE 1231
8441	0034H7	8441AU	J	4100	21			ENV CONT SYS	DBASE 1232
8441	0034H7	8441AV	J	4100	21			MSG DEVELOPER LUKÉ ENV PH 2701	DBASE 1233
8441	0034H7	8441AW	J	4100	21			SGS JAMES EDWARDS ENV	DBASE 1234
8441	0034H7	8441AX	J	4100	21			F15 DATA USED	DBASE 1235
8441	0034H7	8441AY	J	4100	21			ENV CONT SYS F15	DBASE 1236
8441	0034H7	8441AZ	J	27	4100	21			DBASE 1237
8441	0034H7	8441BA	J	4100	21	13	29L	2-22X1 1-31X1	DBASE 1238
8441	0034H7	8441BB	J	4100	21	24	29L	2-22X1	DBASE 1239
8441	0034H7	8441BC	J	47	4100	21	24	29L 2-22X1	DBASE 1240
8441	0034H7	8441BD	J	54	4100	21	27	29L 1-22X1	DBASE 1241
8441	0034H7	8441BE	J	6	4100	21	62	29L 2-21X2	DBASE 1242
8441	0034H7	8441BF	J	4100	23				DBASE 1243
8441	0034H7	8441BG	J	4100	23				DBASE 1244
8441	0034H7	8441BH	J	4100	23			BLEED / SERVICE AIR	DBASE 1245
8441	0034H7	8441BI	J	4100	23	16	29L	1-22X1	DBASE 1246
8441	0034H7	8441BJ	J	4100	21				DBASE 1247
8441	0034H7	8441BK	J	4100	23				DBASE 1248
8441	0034H7	8441BL	J	4100	21			AVN EOP COOLING	DBASE 1249
8441	0034H7	8441BM	J	4100	23	20	29L	1-22X1	DBASE 1250
8441	0034H7	8441BN	J	4100	21				DBASE 1251

SD1A12 L-1442	IO1A02	G	00357	41430	23			DBASE 1252
							CREW COMPT COOL / PRES	DBASE 1253
IO1A12 K-1442		E	75	41430	23	25	29L 2-22X1	DBASE 1254
IO1A12 M-1442		E	75	41430	23	5	29L 2-22X1	DBASE 1255
IO1A12 Q-1442		D		41430	21			DBASE 1256
		M		42100			ELECT PWR	DBASE 1257
		M		42100			F15 M3MA FOR 42404 21737 / 41 FOR	DBASE 1258
		M		42100			42303 422 FOR 42401 41 K 421 FOR	DBASE 1259
		M		42100			42000 423 FOR 42400 / 000	DBASE 1260
		M		42100			A77 42000 3 G 7 9 FOR 42500	DBASE 1261
		M		42100			F15 M3MA 30	DBASE 1262
		M		42100			SSG CLAYTON LUKE ELEC	DBASE 1263
		M		42100			SSG SMITH EDWARD OS ELECT	DBASE 1264
		M		42100			SSG SHAW LUKE PH 2473 ELECT RES	DBASE 1265
		M		42100			ELECT PWR F15	DBASE 1266
02401	Q-2401	E	67	42100	21			DBASE 1267
02401	Q-2401	E	67	42100	21			DBASE 1268
02401	Q-2401	E	67	42100	21			DBASE 1269
02403	R-2403	E	25	42100	21	40	29L 2-21X1 1060	DBASE 1270
02403	R-2403	E	25	42100	21	15	29L 2-31X1	DBASE 1271
02403	R-2403	E	25	42100	21	15	29L 2-31X1	DBASE 1272
02403	R-2403	E	25	42100	21	15	29L 2-31X1	DBASE 1273
02403	R-2403	E	25	42100	21	15	29L 2-31X1	DBASE 1274
02403	R-2403	E	25	42100	21	15	29L 2-31X1	DBASE 1275
02402	SHOP	D		42400	23			DBASE 1276
02402	SHOP	D		42400	23			DBASE 1277
		M		42400			ELECT PWR SYSTEMS	DBASE 1278
IO2401	M-2401	E	50	42400	23	46	29L 2-23X1	DBASE 1279
IO2401	M-2401	E	50	42400	23	46	29L 2-23X1	DBASE 1280
IO2401	M-2401	E	50	42400	23	46	29L 2-23X1	DBASE 1281
IO2401	M-2401	E	50	42400	23	46	29L 2-23X1	DBASE 1282
IO2401	M-2401	E	50	42400	23	46	29L 2-23X1	DBASE 1283
IO2401	M-2401	E	50	42400	23	46	29L 2-23X1	DBASE 1284
		M		44000			EXT LIGHTING	DBASE 1285
		M		44000			F15 442 DATA USED	DBASE 1286
		M		44000			F15 M3MA 44000	DBASE 1287
		M		44000			EXT LIGHTING F15	DBASE 1288
04001	F-4001	E	2	44000	21			DBASE 1289
04001	F-4001	E	2	44000	21			DBASE 1290
04001	F-4001	E	2	44000	21			DBASE 1291
04001	F-4001	E	2	44000	21			DBASE 1292
04001	F-4001	E	2	44000	21			DBASE 1293
04001	F-4001	E	2	44000	21			DBASE 1294
04001	F-4001	E	2	44000	21			DBASE 1295
04001	F-4001	E	2	44000	21			DBASE 1296
04001	F-4001	E	2	44000	21			DBASE 1297
04001	F-4001	E	2	44000	21			DBASE 1298
04001	F-4001	E	2	44000	21			DBASE 1299
04001	F-4001	E	2	44000	21			DBASE 1300
04001	F-4001	E	2	44000	21			DBASE 1301
04001	F-4001	E	2	44000	21			DBASE 1302
04001	F-4001	E	2	44000	21			DBASE 1303
04001	F-4001	E	2	44000	21			DBASE 1304
04001	F-4001	E	2	44000	21			DBASE 1305
04001	F-4001	E	2	44000	21			DBASE 1306
04001	F-4001	E	2	44000	21			DBASE 1307
04001	F-4001	E	2	44000	21			DBASE 1308
04001	F-4001	E	2	44000	21			DBASE 1309
04001	F-4001	E	2	44000	21			DBASE 1310
04001	F-4001	E	2	44000	21			DBASE 1311
04001	F-4001	E	2	44000	21			DBASE 1312
04001	F-4001	E	2	44000	21			DBASE 1313
04001	F-4001	E	2	44000	21			DBASE 1314
04001	F-4001	E	2	44000	21			DBASE 1315
04001	F-4001	E	2	44000	21			DBASE 1316
04001	F-4001	E	2	44000	21			DBASE 1317
04001	F-4001	E	2	44000	21			DBASE 1318
04001	F-4001	E	2	44000	21			DBASE 1319
04001	F-4001	E	2	44000	21			DBASE 1320
04001	F-4001	E	2	44000	21			DBASE 1321
04001	F-4001	E	2	44000	21			DBASE 1322
04001	F-4001	E	2	44000	21			DBASE 1323
04001	F-4001	E	2	44000	21			DBASE 1324
04001	F-4001	E	2	44000	21			DBASE 1325
04001	F-4001	E	2	44000	21			DBASE 1326
04001	F-4001	E	2	44000	21			DBASE 1327
04001	F-4001	E	2	44000	21			DBASE 1328
04001	F-4001	E	2	44000	21			DBASE 1329
04001	F-4001	E	2	44000	21			DBASE 1330
04001	F-4001	E	2	44000	21			DBASE 1331
04001	F-4001	E	2	44000	21			DBASE 1332
04001	F-4001	E	2	44000	21			DBASE 1333
04001	F-4001	E	2	44000	21			DBASE 1334
04001	F-4001	E	2	44000	21			DBASE 1335
04001	F-4001	E	2	44000	21			DBASE 1336
04001	F-4001	E	2	44000	21			DBASE 1337
04001	F-4001	E	2	44000	21			DBASE 1338
04001	F-4001	E	2	44000	21			DBASE 1339
04001	F-4001	E	2	44000	21			DBASE 1340
04001	F-4001	E	2	44000	21			DBASE 1341
04001	F-4001	E	2	44000	21			DBASE 1342
04001	F-4001	E	2	44000	21			DBASE 1343
04001	F-4001	E	2	44000	21			DBASE 1344
04001	F-4001	E	2	44000	21			DBASE 1345
04001	F-4001	E	2	44000	21			DBASE 1346
04001	F-4001	E	2	44000	21			DBASE 1347
04001	F-4001	E	2	44000	21			DBASE 1348
04001	F-4001	E	2	44000	21			DBASE 1349
04001	F-4001	E	2	44000	21			DBASE 1350
04001	F-4001	E	2	44000	21			DBASE 1351
04001	F-4001	E	2	44000	21			DBASE 1352
04001	F-4001	E	2	44000	21			DBASE 1353
04001	F-4001	E	2	44000	21			DBASE 1354
04001	F-4001	E	2	44000	21			DBASE 1355
04001	F-4001	E	2	44000	21			DBASE 1356
04001	F-4001	E	2	44000	21			DBASE 1357
04001	F-4001	E	2	44000	21			DBASE 1358
04001	F-4001	E	2	44000	21			DBASE 1359
04001	F-4001	E	2	44000	21			DBASE 1360
04001	F-4001	E	2	44000	21			DBASE 1361
04001	F-4001	E	2	44000	21			DBASE 1362
04001	F-4001	E	2	44000	21			DBASE 1363
04001	F-4001	E	2	44000	21			DBASE 1364
04001	F-4001	E	2	44000	21			DBASE 1365
04001	F-4001	E	2	44000	21			DBASE 1366
04001	F-4001	E	2	44000	21			DBASE 1367
04001	F-4001	E	2	44000	21			DBASE 1368
04001	F-4001	E	2	44000	21			DBASE 1369
04001	F-4001	E	2	44000	21			DBASE 1370
04001	F-4001	E	2	44000	21			DBASE 1371
04001	F-4001	E	2	44000	21			DBASE 1372
04001	F-4001	E	2	44000	21			DBASE 1373
04001	F-4001	E	2	44000	21			DBASE 1374
04001	F-4001	E	2	44000	21			DBASE 1375
04001	F-4001	E	2	44000	21			DBASE 1376
04001	F-4001	E	2	44000	21			DBASE 1377
04001	F-4001	E	2	44000	21			DBASE 1378
04001	F-4001	E	2	44000	21			DBASE 1379
04001	F-4001	E	2	44000	21			DBASE 1380
04001	F-4001	E	2	44000	21			DBASE 1381
04001	F-4001	E	2	44000	21			DBASE 1382
04001	F-4001	E	2	44000	21			DBASE 1383
04001	F-4001	E	2	44000	21			DBASE 1384
04001	F-4001	E	2	44000	21			DBASE 1385
04001	F-4001	E	2	44000	21			DBASE 1386
04001	F-4001	E	2	44000	21			DBASE 1387
04001	F-4001	E	2	44000	21			DBASE 1388
04001	F-4001	E	2	44000	21			DBASE 1389
04001	F-4001	E	2	44000	21			DBASE 1390
04001	F-4001	E	2	44000	21			DBASE 1391
04001	F-4001	E	2	44000	21			DBASE 1392
04001	F-4001	E	2	44000	21			DBASE 1393
04001	F-4001	E	2	44000	21			DBASE 1394
04001	F-4001	E	2	44000	21			DBASE 1395
04001	F-4001	E	2	44000	21			DBASE 1396
04001	F-4001	E						

	L-5300	4	45000	NO 2 HYD SYS PWR			09ASE 1334
I0501	4-5300	E	35 45000*23	26 29L 2421Y2			09ASE 1335
I0502	4-5300	E	13 45000*23	15 29L 1421Y2			09ASE 1336
I0503	4-5300	E	50 45000*23	15 29L 1421Y2			09ASE 1337
I0504	0-5300	I	45000 21	29 29L 2421Y2			09ASE 1338
			45000	UTILITY HYD SYS			09ASE 1339
		M	45000	F4E 4513 F11F 45A2A 2 FOR 4500E /			09ASE 1390
		M	45000	45000			09ASE 1391
		M	45000	F15 MSBMA 30			09ASE 1392
		M	45000	EXPECTED TO BE WORSE UNLESS AM40 SYS			09ASE 1393
		M	45000	PROBLEMS CORRECTED			09ASE 1394
		M	45000	UTILITY HYD SYS F15			09ASE 1395
05000	F4E000	05000	F	23 45000 21			09ASE 1396
05001	4-5000	E	32 45000*23	26 29L 2421Y2			09ASE 1397
05002	4-5000	E	07 45000*23	15 29L 1421Y2			09ASE 1398
05003	4-5000	E	50 45000*23	15 29L 1421Y2			09ASE 1399
05004	4-5000	E	26 45000*23	34 29L 2421Y2 1TU224			09ASE 1400
05005	4-5000	E	67 45000*23	18 29L 2421Y2 1TU224			09ASE 1401
05006	4-5000	E	02 45000*23	8 29L 2326Y2 1TU224			09ASE 1402
05007	4-5000	E	01 45000*23	7 29L 2326Y2 1TU224			09ASE 1403
05008	4-5000	E	01 45000*23	10 29L 2421Y2 1TU224			09ASE 1404
05009	4-5000	E	02 45000*23	16 29L 2431Y1 1TU224			09ASE 1405
05010	4-5000	E	01 45000*23	10 29L 1531Y2 1TU224			09ASE 1406
05011	4-5000	E	02 45000*23	16 29L 2431Y1 1TU224			09ASE 1407
05012	4-5000	E	02 45000*23	16 29L 2431Y1 1TU224			09ASE 1408
	L-5000	4	45000	UTILITY HYD SYS			09ASE 1409
I05013	4-5000	E	32 45000*23	26 29L 2421Y2			09ASE 1410
I05014	4-5000	E	07 45000*23	15 29L 1421Y2			09ASE 1411
I05015	4-5000	E	50 45000*23	15 29L 1421Y2			09ASE 1412
I05016	4-5000	E	03 45000*23	13 29L 2536Y2			09ASE 1413
I05017	0-5000	I	45000 21	27 29L 2421Y2			09ASE 1414
		4	46400	INTERNAL FUEL SYS			09ASE 1415
		4	46400	F4E 461 DATA USED			09ASE 1416
		4	46400	TSS POOLE LUKE INST			09ASE 1417
		4	46400	MSG WILLIAMS LUKE FUEL			09ASE 1418
		4	46400	SSG PRIN LUKE FUEL			09ASE 1419
		M	46400	F15 MSBMA 30			09ASE 1420
		4	46400	INTERNAL FUEL SYS F15			09ASE 1421
06401	F4E000	06401	F	32 46400 21			09ASE 1422
06402	0U44Y	06402	E	56 46400 21			09ASE 1423
06403	FUEL	06403	J	46400 21	5 29L 2431Y1		09ASE 1424
06404	0FUEL	06404	J	46400 21	20 29L 3431Y1		09ASE 1425
06405	4-6400	E	30 46400*23	120 29L 2424Y1			09ASE 1426
06406	4-6400	E	51 46400*23	67 29L 2424Y1			09ASE 1427
06407	4-6400	E	21 46400*23	720 29L 2424Y1			09ASE 1428
06408	00RNG7	06408	E	44 46400 21			09ASE 1429
06409	4-6400	E	33 46400*23	21 29L 2431Y1			09ASE 1430
06410	4-6400	E	33 46400*23	10 29L 1531Y1			09ASE 1431
06411	4-6400	E	34 46400*23	50 29L 2431Y1			09ASE 1432
06412	0U44Y	06412	J	46400 21			09ASE 1433
06413	4-6400	E	40 46400*23	5 29L 2424Y1 1431Y1			09ASE 1434
06414	4-6400	E	40 46400*23	5 29L 2424Y1 1431Y1			09ASE 1435
06415	4-6400	E	03 46400*23	10 29L 1531Y1			09ASE 1436
06416	4-6400	E	07 46400*23	25 29L 2431Y1			09ASE 1437
06417	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1438
06418	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1439
06419	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1440
06420	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1441
06421	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1442
06422	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1443
06423	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1444
06424	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1445
06425	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1446
06426	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1447
06427	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1448
06428	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1449
06429	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1450
06430	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1451
06431	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1452
06432	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1453
06433	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1454
06434	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1455
06435	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1456
06436	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1457
06437	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1458
06438	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1459
06439	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1460
06440	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1461
06441	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1462
06442	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1463
06443	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1464
06444	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1465
06445	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1466
06446	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1467
06447	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1468
06448	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1469
06449	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1470
06450	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1471
06451	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1472
06452	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1473
06453	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1474
06454	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1475
06455	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1476
06456	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1477
06457	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1478
06458	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1479
06459	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1480
06460	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1481
06461	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1482
06462	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1483
06463	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1484
06464	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1485
06465	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1486
06466	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1487
06467	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1488
06468	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1489
06469	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1490
06470	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1491
06471	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1492
06472	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1493
06473	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1494
06474	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1495
06475	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1496
06476	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1497
06477	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1498
06478	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1499
06479	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1500
06480	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1501
06481	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1502
06482	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1503
06483	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1504
06484	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1505
06485	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1506
06486	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1507
06487	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1508
06488	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1509
06489	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1510
06490	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1511
06491	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1512
06492	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1513
06493	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1514
06494	4-6400	E	01 46400*23	15 29L 2431Y1			09ASE 1515
06495	4-6400	E					

06801	F+6801	06801	F	355	46810	21				06801	1450			
06802	06802	06802	E	40	46810	21				06802	1451			
06803	CALGPJ	06803	C		46810	21				06803	1452			
06804	06804	06804	J		46810	21	45	29L	2424X0	1060	06804	1453		
06805	R+6801	06805	J		46810	21	45	29L	2424X0	1060	06805	1454		
06806	06806	06806	E	3	46810	21	17	29L	5431X1		06806	1455		
06807	06807	06807	E	0	46810	21	18	29L	2432X1		06807	1456		
06808	M+6801	06808	E	15	46810	21	11	29L	1531X0		06808	1457		
06809	06809	06809	E	07	46810	21	13	29L	2531X3		06809	1458		
06810	S+0P	06810	J		46810	23					06810	1459		
06811	L+6801	06811	J		46810	23					06811	1460		
					46810				EXTERNAL WG TANKS		06811	1461		
06812	M+6801	06812	E	21	46810	23	57	29L	2424X0		06812	1462		
06813	K+6801	06813	E	15	46810	23	34	29L	2424X0		06813	1463		
06814	M+6801	06814	E	5	46810	23	16	29L	1424X0		06814	1464		
06815	M+6801	06815	E	03	46810	23	23	29L	2531X0		06815	1465		
06816	M+6802	06816	E	02	46810	23	19	29L	1531X1		06816	1466		
06817	M+6802	06817	E	0	46810	23	21	29L	2531X3		06817	1467		
06818	K+6801	06818	E	14	46810	23	03	29L	2536X0		06818	1468		
06819	06819	06819	J		46810	21					06819	1469		
					46810				REFUEL NTWK		06819	1470		
					46810				F4E 4631 DATA USED		06819	1471		
					46810				F15 MSBMA NO FAILURE		06819	1472		
					46810				REFUEL SYS F15		06819	1473		
06820	F+6801	06820	F	305	46810	21					06820	1474		
06821	06821	06821	J		46810	21					06821	1475		
06822	CALGPJ	06822	C		46810	21					06822	1476		
06823	06823	06823	C		46810	21					06823	1477		
06824	CALTTJ	06824	C		46810	21					06824	1478		
06825	F+6801	06825	E	15	46810	21	16	29L	2423X0	1424X0	1TU228	1060	06825	1479
06826	R+6801	06826	J		46810	21	60	29L	2424X0	1060	1TU228		06826	1480
06827	M+6801	06827	J		46810	21	15	29L	2424X0	1060	1TU228	1431X1	06827	1481
06828	M+6801	06828	E	55	46810	21	16	29L	2424X0	1060	1TU228		06828	1482
06829	M+6801	06829	E	15	46810	21	30	29L	2421X2	1060	1TU228		06829	1483
06830	R+6802	06830	E	14	46810	21	60	29L	2424X0				06830	1484
06831	M+6801	06831	J		46810	21	14	29L	2424X0	1431X1			06831	1485
06832	M+6801	06832	E	01	46810	21	22	29L	2424X0				06832	1486
06833	M+6802	06833	E	3	46810	21	27	29L	2531X3				06833	1487
06834	S+0P	06834	J		46810	23							06834	1488
06835	L+6801	06835	J		46810	23							06835	1489
					46810				AIR REFUEL SYS				06835	1490
06836	M+6801	06836	E	28	46810	23	15	29L	1531X0				06836	1491
06837	M+6801	06837	E	23	46810	23	10	29L	1421X2				06837	1492
06838	M+6801	06838	E	42	46810	23	10	29L	1424X0				06838	1493
06839	06839	06839	J		46810	21							06839	1494
					46810				FUEL CNT & WING INDICATING SYS				06839	1495
					46810				F4E 464				06839	1496
					46810				FUEL CNT & WING INDICATING SYS F15				06839	1497
06840	F+6801	06840	F	32	46810	21							06840	1498
06841	06841	06841	J		46810	21							06841	1499
06842	CALGPJ	06842	C		46810	21							06842	1500
06843	F+6801	06843	J		46810	21	10	29L	2326X2	1060			06843	1501
06844	R+6802	06844	E	97	46810	21	42	29L	2424X0				06844	1502
06845	M+6801	06845	E	3	46810	21	14	29L	2424X0	1431X1			06845	1503
06846	S+0P	06846	J		46810	23							06846	1504
06847	L+6801	06847	J		46810	23							06847	1505
					46810				FUEL CNT & WING IND SYS				06847	1506
06848	M+6801	06848	E	50	46810	23	03	29L	1424X0				06848	1507
06849	K+6801	06849	E	16	46810	23	25	29L	2326X1				06849	1508
06850	M+6801	06850	E	34	46810	23	10	29L	2326X1				06850	1509
06851	06851	06851	J		46810	21							06851	1510
					47400				THE FGL IS LOX NTWK				06851	1511
					47400				F11F 47A DATA USED				06851	1512
					47400				LOX F15				06851	1513
07400	F+7400	07400	F	61	47400	21							07400	1514
07401	R+7400	07401	E	47	47400	21	11	29L	2422X1				07401	1515

07A01	07A01	07A02	E	12	47A00	21	15	29L	132632			08ASE	1516
07A01	07A02	07A02	E	21	47A00	21	19	29L	132631			09ASE	1517
07A01	07A03	07A03	E	3	47A00	21	9	29L	2422X1			08ASE	1519
07A01	07A03	07A03	E	3	47A00	21	10	29L	232692			09ASE	1519
07A03	07A03		O		47A00	21	05	29L	2422X1			09ASE	1521
07A03	07A03		E	04	47A00	21	5	29L	1-31X1			09ASE	1521
07A02	07A00	07A00	O		47A00	23						08ASE	1522
07A02	07A00	07A00	O		47A00	23						08ASE	1523
	L47A00		4		47A00				LCX SYS			09ASE	1524
107A01	47A00		E	11	47A00	23	21	29L	2-22X1			08ASE	1525
107A01	47A00		E	89	47A00	23	17	29L	1-22X1			08ASE	1526
107A01	07A00		I		47A00	21	5	29L	2431X1			08ASE	1527
			4		49A00				FIRE DETECT & WARNING SYS			08ASE	1528
			4		49A00				F11F 43A DATA USED			08ASE	1529
			M		49A00				FIRE DETECT & WARNING ENG & AMMO BAY			08ASE	1530
09A01	09A01	09A01	E	05	49A00	21						09ASE	1531
09A01	09A00	09A02	O		49A00	21	24	29L	2423X0			08ASE	1532
09A02	09A00	09A00	E	46	49A00	21	13	29L	2423X0			08ASE	1533
09A02	09A00	09A00	E	33	49A00	21	09	29L	1423X0			08ASE	1534
09A02	09A00	09A00	E	03	49A00	21	11	29L	1-31X1			08ASE	1535
09A02	09A00	09A00	E	17	49A00	21	06	29L	2-31X1			08ASE	1536
09A03	09A00	09A00	O		49A00	21	08	29L	1-23X0			08ASE	1537
09A04	09A00	09A00	O		49A00	21						08ASE	1538
			4		51A00				THIS IS LAUNCH FOR 51A00 AP58			08ASE	1539
E1A01	E1A00	E1A01	E	15	51A00	21						08ASE	1540
E1A01	E1A00	E1A00	E	10	51A00	21	4	29L	1326L2			08ASE	1541
E1A01	E1A00	E1A00	E	16	51A00	21	2	29L	1326L2			08ASE	1542
E1A01	E1A00	E1A00	E	11	51A00	21	2	29L	1326L2			08ASE	1543
E1A01	E1A00	E1A00	E	14	51A00	21	3	29L	1326L2			08ASE	1544
E1A01	E1A00	E1A00	E	15	51A00	21	3	29L	1326L2			08ASE	1545
E1A01	E1A00	E1A00	E	15	51A00	21	3	29L	1326L2			08ASE	1546
			4		51A00				VERIFIED BY IS POOLE USE F15 DATA			08ASE	1547
			4		51A00				F15 INSTRUMENTS 51A00			08ASE	1548
E1A01	E1A00	E1A00	E	50	51A00	21						08ASE	1549
E1A01	E1A00	E1A00	O		51A00	21						08ASE	1550
E1A02	E1A00	E1A00	O		51A00	21						08ASE	1551
E1A02	E1A00	E1A00	E	75	51A00	21	15	29L	232692 1060 105			08ASE	1552
E1A02	E1A00	E1A00	E	25	51A00	21	15	29L	232692 1431X1 1050 105			08ASE	1553
E1A03	SHOP	S21A00	O		51A00	23						08ASE	1554
			4		51A00				FOR SEPARATE CAPABILITY TO MARTIN FEELS			08ASE	1555
			M		51A00				AUTOMATIC TEST STATION FEELS			08ASE	1556
			4		51A00				THAT ITEMS THAT ARE NOTS ONLY WILL BE			08ASE	1557
			M		51A00				ABOUT 90 PERCENT N ACTIONS			08ASE	1558
			4		51A00				G NOOP COMP DATA F11F			08ASE	1559
			4		51A00				F15 INSTRUMENT MUST HAVE NO SHOP DATA			08ASE	1560
S21A01	L51A00	I51A00	O	00549	51A00	23						08ASE	1561
I51A01	051A00	I51A01	O		51A00	21	10	29L	232692 1050 105			08ASE	1562
I51A01	051A00	I51A01	O		51A00	23						08ASE	1563
I51A01	051A00	I51A02	O		51A00	23						08ASE	1564
I51A02	051A00	I51A02	O	10	51A00	23	15	29L	132691 1P7S24			08ASE	1565
I51A02	051A00	I51A02	O	90	51A00	23	5	29L	132691 1P7S24			08ASE	1566
S21A01	L51A00	I51A00	O	00238	51A00	23						08ASE	1567
			4		51A00				F15 AIRSPEED IND 51A00 G FROM F15			08ASE	1568
I51A00	051A00	I51A00	O	10	51A00	23	22	29L	132691			08ASE	1569
I51A00	051A00	I51A00	O	90	51A00	23	10	29L	132691			08ASE	1570
I51A00	051A00	I51A00	O		51A00	21	10	29L	232692 1060 105			08ASE	1571
S21A01	L51A00	I51A00	O	0001F	51A00	23						08ASE	1572
			4		51A00				F15 VERTICAL SPEED IND 51A00 G NO DATA			08ASE	1573
I51A00	051A00	I51A00	O	10	51A00	23	10	29L	132691			08ASE	1574
I51A00	051A00	I51A00	O	90	51A00	23	10	29L	132691			08ASE	1575
I51A00	051A00	I51A00	O		51A00	21	10	29L	232692 1060 105			08ASE	1576
S21A01	L51A00	I51A00	O	00119	51A00	23						08ASE	1577
			4		51A00				F15 STAY AIRSPEED IND 51A00 G FROM F15			08ASE	1578
I51A00	051A00	I51A00	O		51A00	21	20	29L	232692 1060 105			08ASE	1579
I51A01	051A00	I51A01	O		51A00	23						08ASE	1580
I51A01	051A00	I51A02	O		51A00	23						08ASE	1581

IE1A02	N51A00	E	9	51A00	23	10 29L 1326A1 1PTS2	DBASE 1592	
IE1A02	K51A00	E	1	51A00	23	5 29L 1326A1 1PTS2	DBASE 1593	
SE1A00	L51A00	IE1A00	G	01169	51A00	23	DBASE 1594	
		M		51A00		F15 PRESSURE ALTIMETER 51A00 G-F111F	DBASE 1595	
IE1A40	N51A40	I		51A00	21	20 29L 232692 1000 105	DBASE 1586	
IE1A40	K51A40	IE1A40	D	51A00	*23		DBASE 1597	
IE1A40	N51A40	IE1A40	C	51A00	23		DBASE 1598	
IE1A42	K51A42	E	1	51A00	23	10 29L 1326A1 1PTS1	DBASE 1599	
IE1A42	N51A42	E	9	51A00	23	5 29L 1326A1 1PTS1	DBASE 1590	
SE1A00	L51A00	IE1A00	G	00290	51A00	23	DBASE 1591	
		M		51A00		F15 STRY GYRO INC G FROM F5E	DBASE 1592	
IE1A00	N51A00	I		51A00	21	20 29L 232692 1000 105	DBASE 1597	
IE1A00	K51A00	IE1A00	D	51A00	*23		DBASE 1594	
IE1A00	N51A00	IE1A00	C	51A00	23		DBASE 1595	
IE1A02	K51A02	E	1	51A00	23	30 29L 1326A1 1PTS10	DBASE 1596	
IE1A02	N51A02	E	9	51A00	23	30 29L 1326A1 1PTS10	DBASE 1597	
SE1A00	L51A00	IE1A00	G	00180	51A00	23	DBASE 1593	
		M		51A00		F15 ALTITUDE INC 51A00 G FROM F-E	DBASE 1598	
IE1A00	K51A00	E	1	51A00	*23	15 29L 132691	DBASE 1600	
IE1A00	N51A00	E	9	51A00	*23	10 29L 132691	DBASE 1601	
IE1A00	N51A00	I		51A00	21	10 29L 232692 1000 105	DBASE 1602	
SE1A00	L51A00	IE1A00	G	00110	51A00	23	DBASE 1603	
		M		51A00		F15 ANGLE OF ATTACK INC 51A00 G NO DATA	DBASE 1604	
IE1A00	K51A00	E	1	51A00	*23	15 29L 132691	DBASE 1605	
IE1A00	N51A00	E	9	51A00	*23	10 29L 132691	DBASE 1606	
IE1A00	N51A00	I		51A00	21	10 29L 232692 1000 105	DBASE 1607	
SE1A00	L51A00	IE1A00	G	00197	51A00	23	DBASE 1608	
		M		51A00		F15 ACCELEROMETER INC 51A00 G FROM F4E	DBASE 1609	
IE1A40	N51A40	I		51A00	21	10 29L 232692 1000 105	DBASE 1610	
IE1A40	K51A40	IE1A40	D	51A00	*23		DBASE 1611	
IE1A40	N51A40	IE1A40	C	51A00	23		DBASE 1612	
IE1A42	K51A42	E	1	51A00	23	10 29L 1326A1 1PTS26	DBASE 1613	
IE1A42	N51A42	E	9	51A00	23	5 29L 1326A1 1PTS26	DBASE 1614	
		M		51E00		F111F AND F4E DATA USED	DBASE 1615	
		M		51E00		VERIFIED BY TS POOLE IF PROBE FIXED	DBASE 1616	
		M		51E00		F15 AIR DATA SYS 51E00 F15=44	DBASE 1617	
E1E00	F51E00	E1E00	F	51	51E00	21	DBASE 1618	
E1E00	OPMG7	E1E00	D		51E00	21	DBASE 1619	
E1E00	CALGPJ	E1E00	C		51E00	21	DBASE 1620	
E1E02	K51E02	E	6	51E00	21	15 29L 232692 1000 105	DBASE 1621	
E1E02	N51E02	E	4	51E00	21	10 29L 1431X1 1000 105	DBASE 1622	
E1E02	SHOP	SE1E00	D		51E00	23	DBASE 1623	
SE1E00	L51E00	IE1E00	G	01190	51E00	23	DBASE 1624	
		M		51E00		USED FOR G SELECT WORK AROUND	DBASE 1625	
		M		51E00		F111F 52944 FOR E SEL AND TIMES	DBASE 1626	
		M		51E00		F15 AIR DATA COMPUTER 51E00 F15 DATA	DBASE 1627	
IE1E00	N51E00	I		51E00	21	20 29L 232692 1000 105	DBASE 1628	
IE1E00	K51E00	IE1E00	D	51E00	*23		DBASE 1629	
IE1E00	N51E00	IE1E00	C	51E00	23		DBASE 1630	
IE1E02	N51E02	E	42	51E00	23	10 29L 1326A1 1PTS25 1TSC	DBASE 1631	
IE1E02	K51E02	E	8	51E00	23	45 29L 1326A1 1PTS25 1TSC	DBASE 1632	
IE1E02	N51E02	E	49	51E00	23	89 29L 1326A1 1PTS25 1TSC	DBASE 1633	
SE1E00	L51E00	IE1E00	G	00250	51E00	23	DBASE 1634	
		M		51E00		F4E 51300 DATA USED	DBASE 1635	
		M		51E00		F15 ANGLE OF ATTACK TRANS 51E00	DBASE 1636	
IE1E00	K51E00	E	10	51E00	*23	15 29L 132691	DBASE 1637	
IE1E00	N51E00	E	9	51E00	*23	10 29L 132691	DBASE 1638	
IE1E00	N51E00	I		51E00	21	20 29L 232692 1000 105	DBASE 1639	
		M		51M00		VERIFIED BY TS POOLE SUGGEST NO SWING	DBASE 1640	
		M		51M00		F15 STRY COMPASS 51M00 F111F USED	DBASE 1641	
E1M00	F51M00	E1M00	F	42	51M00	21	DBASE 1642	
E1M00	OPMG7	E1M00	D		51M00	21	DBASE 1643	
E1M00	CALGPJ	E1M00	C		51M00	21	DBASE 1644	
E1M02	K51M02	E1M02	E	60	51M00	21	40 29L 232692 1000 105	DBASE 1645
E1M02	N51M02	E1M02	E	40	51M00	21	10 29L 1431X1 1000 105	DBASE 1646
E1M02	SHOP	SE1M00	D		51M00	23	DBASE 1647	

SE1M30	LS1M80	IE1M90	D	51M00	23					DBASE	1648		
				51M00						DBASE	1649		
				51M00						DBASE	1650		
IE1M30	OS1M80		I	51M00	21	40	29L	232682	1060	105	DBASE	1651	
IE1M30	OS1M80		D	51M00	*23						DBASE	1652	
				51M00							DBASE	1653	
				51M00							DBASE	1654	
E1M01	OS1M80	E1M01	F	27	51M00	21					DBASE	1655	
E1M01	OS1M80	E1M01	D		51M00	21					DBASE	1656	
E1M01	OS1M80	E1M01	D		51M00	21					DBASE	1657	
E1M02	OS1M80	E1M02	E	29	51M00	21	20	29L	232682	1060	105	DBASE	1658
E1M02	OS1M80	E1M02	E	13	51M00	21	13	29L	1-31X1	1060	105	DBASE	1659
E1M02	OS1M80	E1M02	E	52	51M00	21	10	29L	232682	1060	105	DBASE	1660
E1M03	OS1M80	E1M03	D		51M00	23					DBASE	1661	
SE1M30	LS1M80	IE1M90	G	01:59	51M00	23					DBASE	1662	
				51M00							DBASE	1663	
				51M00							DBASE	1664	
IE1M01	OS1M80		I	51M00	21	15	29L	232682	1060	105	DBASE	1665	
IE1M01	OS1M80	IE1M01	D		51M00	*23					DBASE	1666	
IE1M01	OS1M80	IE1M01	D		51M00	23					DBASE	1667	
IE1M02	OS1M80		E	10	51M00	23	20	29L	132641	1PTS40	1TSC	DBASE	1668
IE1M02	OS1M80		E	01	51M00	23	20	29L	132641	1PTS40	1TSC	DBASE	1669
SE1M30	LS1M80	IE1M90	G	03:59	51M00	23					DBASE	1670	
				51M00							DBASE	1671	
				51M00							DBASE	1672	
IE1M01	OS1M80		I	51M00	21	15	29L	232682	1060	105	DBASE	1673	
IE1M01	OS1M80	IE1M01	D		51M00	*23					DBASE	1674	
IE1M01	OS1M80	IE1M01	D		51M00	23					DBASE	1675	
IE1M01	OS1M80	IE1M01	D		51M00	23					DBASE	1676	
IE1M02	OS1M80		E	1	51M00	23	25	29L	132641	1PTS40	1TSC	DBASE	1677
IE1M02	OS1M80		E	01	51M00	23	26	29L	132641	1PTS40	1TSC	DBASE	1678
				52A00							DBASE	1679	
				52A00							DBASE	1680	
				52A00							DBASE	1681	
E2A01	OS2A00	E2A01	F	35	52A00	21					DBASE	1682	
E2A01	OS2A00	E2A01	D		52A00	21					DBASE	1683	
E2A01	OS2A00	E2A01	D		52A00	21					DBASE	1684	
E2A02	OS2A00	E2A02	E	74	52A00	21	24	29L	232682	1060	1TU228	DBASE	1685
E2A02	OS2A00	E2A02	E	5	52A00	21	10	29L	1-31X1			DBASE	1686
E2A02	OS2A00	E2A02	E	12	52A00	21	15	29L	1-31X1	1060	1TU228	DBASE	1687
E2A02	OS2A00	E2A02	E	17	52A00	21	17	29L	232682	1060	1TU228	DBASE	1688
E2A03	OS2A00	E2A03	D		52A00	23					DBASE	1689	
SE2A01	LS2A00	IE2A01	G	00:10	52A00	23					DBASE	1690	
				52A00							DBASE	1691	
				52A00							DBASE	1692	
				52A00							DBASE	1693	
				52A00							DBASE	1694	
IE2A01	OS2A00		I	52A00	21	25	29L	232682	1060	1TU228	DBASE	1695	
IE2A01	OS2A00	IE2A01	D		52A00	*23					DBASE	1696	
IE2A01	OS2A00	IE2A01	D		52A00	23					DBASE	1697	
IE2A02	OS2A00		E	33	52A00	23	23	29L	132641	1PTS12	1TSC	DBASE	1698
IE2A02	OS2A00		E	34	52A00	23	25	29L	132641	1PTS12	1TSC	DBASE	1699
IE2A02	OS2A00		E	33	52A00	23	40	29L	132641	1PTS12	1TSC	DBASE	1700
SE2A01	LS2A00	IE2A01	G	00:23	52A00	23					DBASE	1701	
				52A00							DBASE	1702	
				52A00							DBASE	1703	
IE2A01	OS2A00		I	52A00	21	25	29L	232682	1060	1TU228	DBASE	1704	
IE2A01	OS2A00	IE2A01	D		52A00	*23					DBASE	1705	
IE2A01	OS2A00	IE2A01	D		52A00	23					DBASE	1706	
IE2A02	OS2A00		E	37	52A00	23	40	29L	132641	1PTS12	1TSC	DBASE	1707
IE2A02	OS2A00		E	33	52A00	23	23	29L	132641	1PTS12	1TSC	DBASE	1708
IE2A02	OS2A00		E	34	52A00	23	25	29L	132641	1PTS12	1TSC	DBASE	1709
SE2A01	LS2A00	IE2A01	G	01:09	52A00	23					DBASE	1710	
				52A00							DBASE	1711	
				52A00							DBASE	1712	
				52A00							DBASE	1713	

			M	52A00	F15 RATE SENSOR ASSY 52A00 G=F111F				ORASE 1714
IE2A7	052A0		I	52A00 21	25 29L 132692 1000	1TU228			ORASE 1715
IE2A00	052A00	IE2A01	D	52A00*23					ORASE 1716
IE2A01	052A00	IE2A02	D	52A00 23					ORASE 1717
IE2A02	052A00		E	43 52A00 23	20 29L 1326A1 1PTS4	1TSC			ORASE 1718
IE2A02	052A00		E	37 52A00 23	20 29L 1326A1 1PTS4	1TSC			ORASE 1719
IE2A02	052A00		E	27 52A00 23	20 29L 1326A1 1PTS4	1TSC			ORASE 1720
SE2A01	L52A01	IE2A01	G	0014 52A00 23					ORASE 1721
			M	52A00	TS ROUCHARD GAVE TIME ESTIMATES G=F111F				ORASE 1722
			M	52A00	F15 ACCELEROMETER SENSOR ASSY 52A00				ORASE 1723
IE2A00	052A00		I	52A00 21	25 29L 132692 1000	1TU228			ORASE 1724
IE2A00	052A00	IE2A01	D	52A00*23					ORASE 1725
IE2A01	052A00	IE2A02	D	52A00 23					ORASE 1726
IE2A02	052A00		E	07 52A00 23	20 29L 1326A1 1PTS27	1TSC			ORASE 1727
IE2A02	052A00		E	21 52A00 23	20 29L 1326A1 1PTS27	1TSC			ORASE 1728
IE2A02	052A00		E	77 52A00 23	20 29L 1326A1 1PTS27	1TSC			ORASE 1729
SE2A01	L52A01	IE2A01	G	0011 52A00 23					ORASE 1730
			M	52A00	TS ROUCHARD COULD NOT EST-NO W TASK				ORASE 1731
			M	52A00	F15 DYNAMIC PRESSURE SENSOR 52A00				ORASE 1732
IE2A40	052A40		I	52A00 21	39 29L 132692 1000	1TU228			ORASE 1733
IE2A40	052A40	IE2A41	D	52A00*23					ORASE 1734
IE2A41	052A40	IE2A42	D	52A00 23					ORASE 1735
IE2A42	052A40		E	17 52A00 23	5 29L 1326A1 1PTS17	1TSC			ORASE 1736
IE2A42	052A40		E	97 52A00 23	5 29L 1326A1 1PTS17	1TSC			ORASE 1737
			M	55L00	THIS IS LAUNCH FOR 55L00 AREA				ORASE 1738
ESL01	055L01	ESL01	F	42 55L00 21					ORASE 1739
ESL01	055L01	ESL02	D	55L00 21	3 29L 1326L2				ORASE 1740
			M	55A00	THERE IS NO COMP EQUIP F15 DATA USED				ORASE 1741
			M	55A00	THIS IS NETWORK FOR BIT ON F15 55A00				ORASE 1742
ES401	055A01	ES401	F	237 55A00 21					ORASE 1743
ES401	055A01	ES401	D	55A00 21					ORASE 1744
ES401	055A01	ES402	D	55A00 21					ORASE 1745
ES402	055A01	ES403	E	05 55A00 21	20 29L 232632 1000	105			ORASE 1746
ES402	055A01	ES403	E	5 55A00 21	20 29L 232692 1000	105			ORASE 1747
ES403	055A01	ES403	D	55A00 21					ORASE 1748
SE5A01	L55A01	IE5A01	G	0011 55A00 23					ORASE 1749
			M	55A00	VERIFIED BY TS OBERLY				ORASE 1750
			M	55A00	F15 ASP 55A00				ORASE 1751
IE5A01	055A01		E	15 55A00*23	25 29L 132691				ORASE 1752
IE5A01	055A01		E	97 55A00*23	25 29L 132691				ORASE 1753
IE5A01	055A01		E	55A00 21	20 29L 232632 1000	105			ORASE 1754
SE5A01	L55A01	IE5A01	G	0011 55A00 23					ORASE 1755
			M	55A00	VERIFIED BY TS OBERLY				ORASE 1756
			M	55A00	F15 BIT CONT 55A00				ORASE 1757
IE5A01	055A01		E	65 55A00*23	40 29L 132681				ORASE 1758
IE5A01	055A01		E	1 55A00*23	30 29L 132681				ORASE 1759
IE5A01	055A01		E	25 55A00*23	30 29L 132681				ORASE 1760
IE5A01	055A01		E	55A00 21	10 29L 232692 1000	105			ORASE 1761
			M	55B00	F111F 55A01 AND 55A01 ARE USED AS COMP DATA				ORASE 1762
			M	55A00	THIS IS NETWORK FOR F15 SIG DATA RECORDER				ORASE 1763
ES301	055B01	ES301	F	65 55B00 21					ORASE 1764
ES301	055B01	ES301	D	55B00 21					ORASE 1765
ES301	055B01	ES302	D	55B00 21					ORASE 1766
ES302	055B01	ES303	D	55B00 21	10 29L 232632 1000	105			ORASE 1767
ES303	055B01	ES303	D	55B00 23					ORASE 1768
SE5B01	L55B01	IE5B01	G	00357 55B00 23					ORASE 1769
			M	55B00	SELECT AND TASK TIME ESTIMATES GIVEN				ORASE 1770
			M	55B00	BY TSgt ROUCHARD CHECK F15 DATA				ORASE 1771
			M	55B00	F15 SIG DATA RECORDER FROM F15 DATA				ORASE 1772
IE5B01	055B01		I	55B00 21	240 29L 232692				ORASE 1773
IE5B01	055B01	IE5B01	D	55B00*23					ORASE 1774
IE5B01	055B01	IE5B02	D	55B00 23					ORASE 1775
IE5B02	055B01		E	30 55B00 23	20 29L 2326A1 1PTS3	1TSC			ORASE 1776
IE5B02	055B01		E	30 55B00 23	05 29L 2326A1 1PTS3	1TSC			ORASE 1777
IE5B02	055B01		E	47 55B00 23	05 29L 2326A1 1PTS3	1TSC			ORASE 1778
SE5B01	L55B01	IE5B01	G	00366 55B00 23					ORASE 1779

	4		55900	VERIFIED BY TS BOUCHARD								DBASE 1780
	M		55900	F15 CASSETTE FROM F111F 55AAA DATA								DBASE 1781
IES9E1 0559E3	I		55900 21	240 29L 232692								DBASE 1782
IES9E1 0559E3	I		55900*23									DBASE 1783
	4		55900	F111F 55A03 USED F15 MS9MA=213								DBASE 1790
	4		55900	VERIFIED BY TS POOLE 2733								DBASE 1795
	4		55900	F15 ACCELEROMETER COUNTER 55003								DBASE 1746
ESC01 055007	E	75	55000 21									DBASE 1767
ESC01 055007	E		55000 21									DBASE 1739
ESC02 055007	E		55000 21									DBASE 1742
ESC02 055007	E	97	55000 21	10 29L 232692 1060 105								DBASE 1791
ESC02 055007	E		55000 21	F SELECT FROM F15 DATA								DBASE 1791
ESC02 055007	E	3	55000 21	10 29L 232692 1060 105								DBASE 1792
ESC02 055007	E		55000 21									DBASE 1793
SE5006 055007	S	0337	55000 23									DBASE 1794
	4		55000	F15 DATA USED NO F111F SHOP DATA								DBASE 1795
	4		55000	TASK TIME EST BY TS BOUCHARD								DBASE 1796
	4		55000	DIGITAL READOUT ELECT COUNTER 55003								DBASE 1797
IES004 055007	I		55000 21	25 29L 232692 1060 105								DBASE 1793
IES004 055007	I		55000*23									DBASE 1799
IES004 055007	I		55000 23									DBASE 1800
IES002 055007	E	6	55000 23	40 29L 132641 1PTS19 1TSC								DBASE 1801
IES002 055007	E	2	55000 23	25 29L 132641 1PTS19 1TSC								DBASE 1812
IES002 055007	E	97	55000 23	25 29L 132641 1PTS19 1TSC								DBASE 1803
SE5006 055007	S	01119	55000 23									DBASE 1804
	4		55000	F15 DATA USED NO F111F DATA AVAIL								DBASE 1805
	M		55000	NO M TASK TIME EST BY TS BOUCHARD								DBASE 1806
	M		55000	LINEAR ELECT ACCELEROMETER 55003								DBASE 1807
IES003 055007	I		55000 21	20 29L 232692 1060 105								DBASE 1808
IES003 055007	I		55000*23									DBASE 1809
IES001 055007	I		55000 23									DBASE 1810
IES002 055007	E	85	55000 23	12 29L 132641 1PTS16 1TSC								DBASE 1811
IES002 055007	E	15	55000 23	10 29L 132641 1PTS16 1TSC								DBASE 1812
	4		57000	THIS IS LAUNCH FOR 57000 AREA								DBASE 1813
E7L01 057007	E	70	57000 21									DBASE 1814
E7L01 057007	E		57000 21	2 29L 132642								DBASE 1815
	4		57000	THIS IS NETWORK FOR F15 CENTRAL								DBASE 1816
	4		57000	COMPUTER 57000 THERE IS NO COMP								DBASE 1817
	4		57000	DATA F15 DATA USED								DBASE 1818
	4		57000	VERIFIED BY TS FAY								DBASE 1819
	M		57000	F15 CENTRAL COMPUTER								DBASE 1820
E7A01 057007	E	20	57000 21									DBASE 1821
E7A01 057007	E		57000 21									DBASE 1822
E7A01 057007	E		57000 21									DBASE 1823
E7A02 057007	E	17	57000 21	8 29L 232642 1060 105 1431X1							DBASE 1824	
E7A02 057007	E	83	57000 21	10 29L 232642 1060 105								DBASE 1825
E7A03 057007	E		57000 23									DBASE 1826
SE7007 057007	S		57000 23									DBASE 1827
	4		57000	WORK AROUND								DBASE 1828
	4		57000	F15 57000 CENT COMP F15 DATA USED								DBASE 1829
IES7001 057007	I		57000 21	8 29L 232642								DBASE 1830
IES7001 057007	I		57000*23									DBASE 1831
IES7001 057007	I		57000 23									DBASE 1832
IES7002 057007	E	65	57000 23	45 29L 132641 1PTS20 1TSC								DBASE 1833
IES7002 057007	E	29	57000 23	35 29L 132641 1PTS20 1TSC								DBASE 1834
IES7002 057007	E	6	57000 23	25 29L 132641 1PTS20 1TSC								DBASE 1835
	4		63000	THIS IS LAUNCH FOR 63000 AREA								DBASE 1836
F3L01 063007	F	12	63000 21									DBASE 1837
F3L01 063007	F	39	63000 21	1 29L 132642								DBASE 1838
F3L01 063007	F	01	63000 21	2 29L 132642								DBASE 1839
F3L01 063007	F	02	63000 21	1 29L 132642								DBASE 1840
F3L01 063007	F	03	63000 21	3 29L 132642								DBASE 1841
	4		63000	COMP DATA FROM F111F AND A70 USED								DBASE 1842
	M		63000	VERIFIED BY TSGT FIELD 7305								DBASE 1843
	M		63000	THIS NETWORK IS FOR THE 63000 AREA F15								DBASE 1844
	M		63000	UNF COMM SET AND 109 F15								DBASE 1845

F3A11	F63A11	F3A11	F	27	63A00	21							DBASE	1346
F3A12	F63A12	F3A12	D		63A00	21							DBASE	1347
F3A14	CALGRP	F3A02	D		63A00	21							DBASE	1348
F3A12	M63A01	F3A03	E	58	63A00	21	13	29L	232602	1050	105		DBASE	1349
F3A12	M63A01	F3A03	E	48	63A00	21	12	29L	232602	1060	105		DBASE	1350
F3A23	SMOP	SF3A11	D		63A00	23							DBASE	1351
SF3A11	L63A01	IF3A01	G	2235	63A00	23							DBASE	1352
					63A00				COMP F111F 63A00				DBASE	1353
					63A00				F15 UHF RT 63A00				DBASE	1354
IF3A01	M63A01	IF3A01	E	77	63A00	23	51	29L	132691				DBASE	1355
IF3A01	K63A01		E	10	63A00	23	30	29L	132691				DBASE	1356
IF3A01	M63A01		E	17	63A00	23	20	29L	132691				DBASE	1357
IF3A40	Q63A02		I		63A00	21	6	29L	232602				DBASE	1358
					63A00				M63A1 INDICATES LMS LEAK CHECK				DBASE	1359
IF3A41	M63A01		D		63A00	23	40		G				DBASE	1360
SF3A11	L63A01	IF3A01	G	2257	63A00	23							DBASE	1361
					63A00				COMP A7D 63B00				DBASE	1362
					63A00				M63A01 DATA REFLECTED AT PERCENT FIX				DBASE	1363
					63A00				EXPERIENCE AND CAPABILITY WILL MAKE A				DBASE	1364
					63A00				65 PERCENT REPAIR POSSIBLE				DBASE	1365
					63A00				F15 UHF PACKUP RECEIVER				DBASE	1366
IF3A01	M63A01		E	89	63A00	23	30	29L	132691				DBASE	1367
IF3A01	K63A01		E	15	63A00	23	20	29L	132691				DBASE	1368
IF3A01	Q63A02		I		63A00	21	8	29L	232602				DBASE	1369
					63A00				THERE IS NO DATA FOR SECURE VOICE OR				DBASE	1370
					63A00				IN SMOP ANTENNA REPAIR				DBASE	1371
					63A00				THIS NETWORK IS FOR THE 63B AREA F15				DBASE	1372
					63A00				IT COVERS ALL CONTROL PANELS				DBASE	1373
					63B00				CONF DATA FROM F111F AND F48				DBASE	1374
					63B00				VERIFIED BY TEST FIELD 7312				DBASE	1375
					63B00				INTEG ONI CONT SET F15				DBASE	1376
F3B11	F63B11	F3B01	F	25	63B00	21							DBASE	1377
F3B12	DC-MG7	F3B01	D		63B00	21							DBASE	1378
F3B14	CALGRP	F3B02	D		63B00	21							DBASE	1379
F3B12	M63B01	F3B03	E	58	63B00	21	25	29L	232602	1050	105		DBASE	1380
F3B12	M63B01	F3B03	E	48	63B00	21	16	29L	232602	1060	105		DBASE	1381
F3B02	M63B01		I	15	63B00	21	10	29L	243141				DBASE	1382
F3B13	S4OP	SF3B01	D		63B00	23							DBASE	1383
SF3B01	L63B01	IF3B01	G	2203	63B00	23							DBASE	1384
					63B00				63B AREA FOR S4OP WORK VERIFIED BY				DBASE	1385
					63B00				TEST OFFICER 7312				DBASE	1386
					63B00				ADDED M63B00 AND ADJ E SELECT MODES				DBASE	1387
					63B00								DBASE	1388
IF3B01	M63B01		E	71	63B00	23	60	29L	132691				DBASE	1389
IF3B01	K63B01		E	15	63B00	23	30	29L	132691				DBASE	1390
IF3B01	M63B01		E	15	63B00	23	30	29L	132691				DBASE	1391
IF3B01	Q63B02		I		63B00	21	20	29L	232602	1060	105		DBASE	1392
SF3B01	L63B01	IF3B01	G	2202	63B00	23							DBASE	1393
					63B00				ADDED M63B00 AND ADJ E SELECT MODES				DBASE	1394
					63B00								DBASE	1395
IF3B01	M63B01		E	75	63B00	23	15	29L	132691				DBASE	1396
IF3B01	K63B01		E	15	63B00	23	10	29L	132691				DBASE	1397
IF3B01	M63B01		E	10	63B00	23	10	29L	132691				DBASE	1398
IF3B01	Q63B02		I		63B00	21	10	29L	232602				DBASE	1399
					63B00				F15 ONI CONTROL USED M63B00 AND 64000				DBASE	1400
					63B00				FROM F111F AS COMP				DBASE	1401
					63B00				F15 NAVATION CONT USED 7100A AND 7100C				DBASE	1402
					63B00				FROM F111F AS COMP				DBASE	1403
					63B00								DBASE	1404
					63B00								DBASE	1405
SF3B01	L63B01	IF3B01	G	22169	63B00	23							DBASE	1406
					63B00				F15 IFF USED 64000 FROM F111F AS COMP				DBASE	1407
IF3B01	M63B01		E	64	63B00	23	15	29L	132691				DBASE	1408
IF3B01	K63B01		E	15	63B00	23	10	29L	132691				DBASE	1409
IF3B01	M63B01		E	16	63B00	23	10	29L	132691				DBASE	1410
IF3B01	Q63B02		I		63B00	21	12	29L	232602				DBASE	1411
SF3B01	L63B01	IF3B01	G	22267	63B00	23							DBASE	1411

			4	63900	F15 MAIN COM1 CONTROL USED 63900 AND	09ASE 1912
			M	63900	ADDED 63900 AND ADJ E SELECT MODES	09ASE 1913
			M	63900	63900 FROM F111F AS COMP	09ASE 1914
			M	63900		09ASE 1915
IF3940	463940	E	55	63900*23	15 29L 132631	09ASE 1916
IF3940	463940	E	05	63900*23	10 29L 132631	09ASE 1917
IF3940	463940	E	10	63900*23	10 29L 132631	09ASE 1918
IF3940	463940	E	10	63900*23	10 29L 132631	09ASE 1919
SF3900	463900	G	0595	63900 23		09ASE 1920
			M	63900	ADDED 63900 AND 63900 AND ADJ SELECT	09ASE 1921
			M	63900	MODES	09ASE 1922
			M	63900		09ASE 1923
IF3900	463900	E	80	63900*23	15 29L 132631	09ASE 1924
IF3900	463900	E	10	63900*23	10 29L 132631	09ASE 1925
IF3900	463900	E	10	63900*23	10 29L 132631	09ASE 1926
IF3900	463900	E	10	63900*23	10 29L 132631	09ASE 1927
			M	63800	THE AAI CONT BOX IS F15 DATA USED	09ASE 1928
			M	65100	THIS IS LAUNCH ERR 65100 AREA	09ASE 1929
F5100	F55100	F	69	65100 21		09ASE 1930
F5100	F55100	F	69	65100 21	2 29L 232612	09ASE 1931
			M	65400	F111F 65400-65400 IS USED AS COMP	09ASE 1932
			M	65400	VERIFIED BY TSOT FIELD	09ASE 1933
			M	65400	THIS IS IFF TRANSPONDER APX101 FOR F15	09ASE 1934
F5100	F55100	F	69	65400 21		09ASE 1935
F5100	F55100	F	69	65400 21		09ASE 1936
F5100	F55100	F	69	65400 21		09ASE 1937
F5100	F55100	F	69	65400 21	18 29L 232602 1050 105	09ASE 1938
F5100	F55100	F	69	65400 21	18 29L 232602 1050 105	09ASE 1939
F5100	F55100	F	69	65400 21	20 29L 143141	09ASE 1940
F5100	F55100	F	69	65400 21		09ASE 1941
SF5100	465100	G	0595	65400 23		09ASE 1942
			M	65400	THERE IS NO DATA TO INDICATE 65400	09ASE 1943
			M	65400	WORK EITHER F15 OR COMP ACFT	09ASE 1944
			M	65400	VERIFIED S40P WORK WITH TSOT OVERLEY	09ASE 1945
			M	65400	PH#7302	09ASE 1946
			M	65400	F15 BY/TX F111F 65400 AS COMP	09ASE 1947
IF5100	465100	E	70	65400*23	16L 29L 132631	09ASE 1948
IF5100	465100	E	5	65400*23	70 29L 132631	09ASE 1949
IF5100	465100	E	25	65400*23	60 29L 132631	09ASE 1950
IF5100	465100	E	15	65400*23	15 29L 232602	09ASE 1951
			M	65900	THIS IS NETWORK FOR AAI IFF F15 DATA	09ASE 1952
			M	65900	USED AS COMP DATA NOT AVAILABLE	09ASE 1953
			M	65900	AAI IFF	09ASE 1954
F5100	F55100	F	69	65900 21		09ASE 1955
F5100	F55100	F	69	65900 21		09ASE 1956
F5100	F55100	F	69	65900 21		09ASE 1957
F5100	F55100	F	69	65900 21	23 29L 232602 1050 105	09ASE 1958
F5100	F55100	F	69	65900 21	7 29L 232602 1050 105	09ASE 1959
F5100	F55100	F	69	65900 21	17 29L 2-3141	09ASE 1960
F5100	F55100	F	69	65900 21	10 29L 232602 1050 105	09ASE 1961
F5100	F55100	F	69	65900 23		09ASE 1962
SF5100	465100	G	0595	65900 23		09ASE 1963
			M	65900	F15 AAI RTA60/APX75 F15 DATA USED	09ASE 1964
			M	65900	ADDED 65900 AND ADJ E SELECT MODES	09ASE 1965
			M	65900	VERIFIED S40P WORK WITH TSOT OVERLEY	09ASE 1966
			M	65900	PH#7302	09ASE 1967
			M	65900	SGT OVERLEY GAVE BEST ESTIMATES OF	09ASE 1968
			M	65900	TASK TIME REQUIREMENTS ALL CREW STRES	09ASE 1969
			M	65900	ARE 1	09ASE 1970
			M	65900	F15 AAI RTA60/APX75	09ASE 1971
IF5100	465100	E	71	65900*23	100 29L 132631	09ASE 1972
IF5100	465100	E	5	65900*23	45 29L 132631	09ASE 1973
IF5100	465100	E	24	65900*23	50 29L 132631	09ASE 1974
IF5100	465100	E	23	65900*23	23 29L 232602	09ASE 1975
SF5100	465100	G	0595	65900 23		09ASE 1976
			M	65900	F15 AAI TARGET PROCESSOR F15 DATA USED	09ASE 1977

IF5E4	0658M3	I		65900	21		23 29L 232602			ORASE	1978
IF594	0659M7	IF59M1	7	65900	23					ORASE	1979
IF5941	0659M7	IF59M2	7	65900	23					ORASE	1980
IF5942	0659M7		E	37 65900	23		72 29L 132641 1PT534 1T50			ORASE	1981
IF5942	0659M3		E	67 65900	23		15 29L 232641 1PT534 1T50			ORASE	1982
			H	71000			THIS IS LAUNCH FOR 71000 AFSA			ORASE	1983
G100	F71000	G1001	F	1 71000	21					ORASE	1984
G1001	071007	IG1001	E	14 71000	21		1 29L 232612			ORASE	1985
G1001	071007	IG1002	E	75 71000	21		3 29L 232612			ORASE	1986
G1001	071007	IG1003	E	11 71000	21		3 29L 232612			ORASE	1987
			H	71000			A70 WAS GIVEN AS COMPARABLE BUT THIS			ORASE	1988
			H	71000			SYSTEM SHOULD BE MUCH BETTER THE			ORASE	1989
			H	71000			ENGINEER W/AV PROGRAM COULD NOT ESTIMATE			ORASE	1990
			H	71000			HOW MUCH BETTER COULD YOU A70 ARE			ORASE	1991
			H	71000			73FAC+73000+73F01+73F02+73F03 THE LAST			ORASE	1992
			H	71000			BEING FOR THE INS UNIT F15 71AK1			ORASE	1993
			H	71000			VERIFIED BY TSGT RAY A70 17 F15=1-			ORASE	1994
			H	71000			THIS IS NETWORK FOR F15 INS 71AK1			ORASE	1995
G1A00	F71A00	G1A01	F	13 71A00	21					ORASE	1996
G1A01	071A07	G1A01A	D	71A00	21					ORASE	1997
G1A01A	071A07	G1A02	D	71A00	21					ORASE	1998
G1A01A	071A07	G1A03	D	4 71A00	21		17 29L 232643 1060 105			ORASE	1999
G1A02	071A07		E	16 71A00	21		22 29L 232642 1060 105			ORASE	2000
G1A02	071A07		H	21 71A00	21		29 29L 232641			ORASE	2001
G1A03	S40P	SG1A03	D	71A00	23					ORASE	2002
SG1A03	L71AK0	IG1A03	D	03136	71A00	23				ORASE	2003
			H	71A00			F15 INS UNIT USED A70 73FAC+73F01			ORASE	2004
			H	71A00			73F02+73000 THE F15 HAS EVERYTHING IN			ORASE	2005
			H	71A00			ONE UNIT E SELECT AND TIMES AND			ORASE	2006
			H	71A00			VERIFIED BY TS TOUCHARD WORK AROUND			ORASE	2007
			H	71A00			E SEL AND TIMES FM A70			ORASE	2008
			H	71A00			F15 INS UNIT			ORASE	2009
IG1AE	071AE0		I	71A00	21		10 29L 232642 1060 105			ORASE	2010
IG1AE	071AE0	IG1AE1	D	71A00	23					ORASE	2011
IG1AE1	071AE0	IG1AE2	D	71A00	23					ORASE	2012
IG1AE2	071AE0		E	49 71A00	23		96 29L 132641 1PT59 1T50			ORASE	2013
IG1AE2	071AE0		E	26 71A00	23		75 29L 132641 1PT59 1T50			ORASE	2014
IG1AE2	071AE0		E	26 71A00	23		58 29L 132641 1PT59 1T50			ORASE	2015
SG1A03	L71AK0	IG1A03	D	03136	71A00	23				ORASE	2016
			H	71A00			F15 INS CONT-INS A70 73FAC DATA USED			ORASE	2017
			H	71A00			E SELECT AND TASK TIMES AND DATA			ORASE	2018
			H	71A00			VERIFIED BY TS TOUCHARD WORK AROUND			ORASE	2019
			H	71A00			F15 INS CONT			ORASE	2020
IG1AK	071AK0		I	71A00	21		10 29L 232642 1060 105			ORASE	2021
IG1AK	071AK0	IG1AK1	D	71A00	23					ORASE	2022
IG1AK1	071AK0	IG1AK2	D	71A00	23					ORASE	2023
IG1AK2	071AK0		E	54 71A00	23		46 29L 132641 1PT56 1T50			ORASE	2024
IG1AK2	071AK0		E	21 71A00	23		11 29L 132641 1PT56 1T50			ORASE	2025
IG1AK2	071AK0		E	25 71A00	23		10 29L 132641 1PT56 1T50			ORASE	2026
			H	71A00			F11F 71CA0 DATA USED			ORASE	2027
			H	71A00			VERIFIED BY TSGT FIELD			ORASE	2028
			H	71A00			THIS IS NETWORK FOR F15 ACF 71B00			ORASE	2029
G1B00	F71B00	G1B01	F	305 71B00	21					ORASE	2030
G1B01	071B07	G1B01A	D	71B00	21					ORASE	2031
G1B01A	071B07	G1B02	D	71B00	21					ORASE	2032
G1B02	071B07	G1B03	E	85 71B00	21		25 29L 232602 1060 105			ORASE	2033
G1B02	071B07		E	15 71B00	21		20 29L 232602 1060 105 143141			ORASE	2034
			H	71B00			THERE IS NO SIGNIFICANT SHOW DATA FOR			ORASE	2035
			H	71B00			THIS SUBSYSTEM DATA BY TOUCHARD			ORASE	2036
			H	71B00			THIS IS NETWORK FOR F15 ACF 71B00			ORASE	2037
G1B03	L71B00	IG1B03	D	71B00	23					ORASE	2038
IG1B03	071B00		E	70 71B00	23		30 29L 132641			ORASE	2039
IG1B03	071B00		E	20 71B00	23		20 29L 132641			ORASE	2040
IG1B03	071B00		E	10 71B00	23		20 29L 132641			ORASE	2041
IG1B03	071B00		I	71B00	21		20 29L 232642 1060 105			ORASE	2042
			H	71B00			F11F 71CA0 DATA USED F15 MSB#4 1000			ORASE	2043

	4		71000	VERIFIED BY TSGT FIELD		09ASE 2044
	4		71001	THIS IS NETWORK FOR F15 ILS 71001		09ASE 2045
G1001	F71001	G1001	F	533	71001 21	09ASE 2046
G1001	032457	G1001	D		71001 21	09ASE 2047
G1002	040601	G1002	D		71001 21	09ASE 2048
G1002	071001	G1002	F	31	71001 21 21 29L 232602 1060 105	09ASE 2049
G1002	071001	G1002	E	11	71001 21 11 29L 232602 1060 105 1431X1	09ASE 2050
G1003	040601	G1003	D		71001 21	09ASE 2051
SG1003	L71001	IG1003	D		71001 23	09ASE 2052
	4		71004	F15 ILS RC/PF 71001 USED F111F 71001		09ASE 2053
	4		71005	VERIFIED BY TS OBERLY		09ASE 2054
	4		71006	F15 ILS RC/PF		09ASE 2055
IG1006	071001	IG1006	E	75	71006*23 30 29L 232601	09ASE 2056
IG1006	071001	IG1006	E	15	71006*23 24 29L 232601	09ASE 2057
IG1006	071001	IG1006	E	16	71006*23 20 29L 232601	09ASE 2058
IG1006	071001	IG1006	I		71006 21 11 29L 232602 1060 105	09ASE 2059
	4		71007	F111F 71001 DATA USED F15 0934 76		09ASE 2060
	4		71008	VERIFIED BY TSGT FIELD 463 37 FOR F111F		09ASE 2061
	4		71009	THIS IS NETWORK FOR F15 TACAN 71009		09ASE 2062
G1011	032457	G1011	D	55	71009 21	09ASE 2063
G1011	040601	G1011	D		71009 21	09ASE 2064
G1011	071001	G1011	E	92	71009 21 25 29L 232602 1060 105	09ASE 2065
G1011	071001	G1011	E		71009 21 15 29L 232602 1060 105 1431X1	09ASE 2066
G1013	040601	G1013	D		71009 23	09ASE 2067
SG1013	L71001	IG1013	D		71009 23	09ASE 2068
	4		71010	F15 TACAN RC/PF 71001 F111F 71001 USED		09ASE 2069
	4		71011	F15 TACAN RC/PF		09ASE 2070
IG1011	071001	IG1011	E	80	71011*23 20 29L 232601	09ASE 2071
IG1011	071001	IG1011	E	15	71011*23 15 29L 232601	09ASE 2072
IG1011	071001	IG1011	E	5	71011*23 15 29L 232601	09ASE 2073
IG1011	071001	IG1011	I		71011 21 20 29L 232602 1060 105	09ASE 2074
	4		71F00	THIS IS NETWORK FOR F15 ATTITUDE HEAD		09ASE 2075
	4		71F00	ING REF F111F 510 DATA USED F15 24		09ASE 2076
	4		71F00	VERIFIED BY TS POOLE		09ASE 2077
	4		71F00	ATTITUDE/HEADING REF		09ASE 2078
G1F01	032457	G1F01	F	52	71F00 21	09ASE 2079
G1F01	040601	G1F01	D		71F00 21	09ASE 2080
G1F01	071001	G1F01	E	56	71F00 21 31 29L 232602 1060 105	09ASE 2081
G1F01	071001	G1F01	E		71F00 21 11 29L 232602 1060 105	09ASE 2082
G1F01	071001	G1F01	E	25	71F00 21 18 29L 1431X1	09ASE 2083
G1F03	040601	G1F03	D		71F00 23	09ASE 2084
SG1F03	L71F00	IG1F03	D	03-65	71F00 23	09ASE 2085
	4		71F00	F15 ELEC-CONT AMP 71F00 F111F 51000		09ASE 2086
	4		71F00	DATA USED WORK AROUND		09ASE 2087
	4		71F00	VERIFIED BY TS BOULHARD		09ASE 2088
	4		71F00	F15 ELEC-CONT AMP		09ASE 2089
IG1F00	071F00	IG1F00	E		71F00 21 15 29L 232602 1060 105	09ASE 2090
IG1F01	071F00	IG1F01	E		71F00*23	09ASE 2091
IG1F01	071F00	IG1F01	E		71F00 23	09ASE 2092
IG1F01	071F00	IG1F01	E	55	71F00 23 40 29L 232601 12TS14 1TSC	09ASE 2093
IG1F01	071F00	IG1F01	E	11	71F00 23 30 29L 232601 12TS14 1TSC	09ASE 2094
IG1F01	071F00	IG1F01	E	27	71F00 23 36 29L 232601 12TS14 1TSC	09ASE 2095
SG1F00	L71F00	IG1F00	D	0330	71F00 23	09ASE 2096
	4		71F00	F15 DISPLACEMENT GYRO F111F 51000 DATA USED		09ASE 2097
	4		71F00	USED FOR RATE SELECT AND TASK TIMES		09ASE 2098
	4		71F00	VERIFIED BY TS BOULHARD		09ASE 2099
	4		71F00	F15 DISPLACEMENT GYRO		09ASE 2100
IG1F01	071F00	IG1F01	E		71F00 21 15 29L 232602 1060 105	09ASE 2101
SG1F01	L71F00	IG1F01	D	0339	71F00 23	09ASE 2102
IG1F01	071F00	IG1F01	E		71F00*23	09ASE 2103
IG1F01	071F00	IG1F01	E		71F00 23	09ASE 2104
IG1F01	071F00	IG1F01	E	10	71F00 23 70 29L 232601 12TS14 1TSC	09ASE 2105
IG1F01	071F00	IG1F01	E	90	71F00 23 70 29L 232601 12TS14 1TSC	09ASE 2106
	4		71F00	F15 COMPASS CONT F111F 51000 DATA USED		09ASE 2107

TIME	DESCRIPTION	TIME	TASK TIMES GIVEN	BASE
74500	F15 MAG A71M0TH DETEC	F111F F104D		DBASE 2117
74500	DATA USED FOR RATE E SELECT AND TASK			DBASE 2118
74500	F15 MAG A71M0TH DETEC			DBASE 2119
74500 21	35 29L 232682 1060 105			DBASE 2120
74500 21	36 29L 232682 1060 105			DBASE 2121
74500 23				DBASE 2122
74500 23				DBASE 2123
74500 23	10 29L 132641 1PT57 1T50			DBASE 2124
74500 23	10 29L 132641 1PT57 1T50			DBASE 2125
74500	THIS IS LAUNCH FOR 74500 AREA			DBASE 2126
74500 21				DBASE 2127
74500 21	2 29L 2326L2			DBASE 2128
74500 21	3 29L 2326L2			DBASE 2129
74500 21	3 29L 2326L2			DBASE 2130
74500 21	3 29L 2326L2			DBASE 2131
74500 21	3 29L 2326L2			DBASE 2132
74500	F15 DATA USED E SELECT AND TIMES GIVEN			DBASE 2133
74500	VERIFIED BY TSGT RAY			DBASE 2134
74500	F15 LEAD COMPUTING GYRO 74500			DBASE 2135
74500 21				DBASE 2136
74500 21				DBASE 2137
74500 21				DBASE 2138
74500 21	3 29L 2326A2 1060 105			DBASE 2139
74500 21	20 29L 2431X1			DBASE 2140
74500 21	50 29L 2326A2 1060 105			DBASE 2141
74500 23				DBASE 2142
74500 23				DBASE 2143
74500	F15 LEAD COMP GYRO 74500 E SELECT AND			DBASE 2144
74500	TASK TIMES GIVEN AND VERIFIED BY			DBASE 2145
74500	TS BOUCHARD			DBASE 2146
74500	F15 LEAD COMP GYRO			DBASE 2147
74500 21	10 29L 2326A2 1060 105			DBASE 2148
74500 23				DBASE 2149
74500 23				DBASE 2150
74500 23	50 29L 132641 1PT59 1T50			DBASE 2151
74500 23	30 29L 132641 1PT59 1T50			DBASE 2152
74500 23	30 29L 132641 1PT59 1T50			DBASE 2153
74500	F111F DATA IS USED AS COMPARABLE WUCS			DBASE 2154
74500	73VAC+73JAC+73VCG MAKE UP MOST OF THE			DBASE 2155
74500	RADAR COMPONENTS THE F15 DIGITAL AND			DBASE 2156
74500	ANALOG PROCESSOR ARE UNTOUCHED FOR THIS			DBASE 2157
74500	REASON SPC ENGINEERS ESTIMATED THAT			DBASE 2158
74500	F15 MSBMA WOULD BE 91 PERCENT OF THE			DBASE 2159
74500	F111F DATA F15 MSBMA 7 F111 MSBMA1			DBASE 2160
74500	VERIFIED BY TSGT RAY			DBASE 2161
74500	THIS IS NETWORK FOR F15 RADAR 74500			DBASE 2162
74500 21				DBASE 2163
74500 21				DBASE 2164
74500 21				DBASE 2165
74500 21	25 29L 2326A2 1060 105			DBASE 2166
74500 21	14 29L 2326A2 1060 105			DBASE 2167
74500 21	20 29L 2431X1			DBASE 2168
74500 23				DBASE 2169
74500 23				DBASE 2170
74500	F15 RA TRANSMITTER 74500 USED F111F			DBASE 2171
74500	73VAC AS PUMP'S SELECTION IS 1/3 OF			DBASE 2172
74500	F111F DATA TIMES AND E SELECT GIVEN			DBASE 2173
74500	AND VERIFIED BY TS BOUCHARD			DBASE 2174
74500	F15 RA TRANSMITTER			DBASE 2175

IG4F10	N7-F10	E	52	74F00*23	90	29L	232691		DBASE	2176
IG4F11	K7-F11	E	2	74F10*23	51	29L	232691		DBASE	2177
IG4F11	N7-F11	E	46	74F10*23	52	29L	232691		DBASE	2178
IG4F11	N7-F11	E		74F00 21	30	29L	232692 1060	105	DBASE	2179
SG4F10	L7-F10	IG4F01	G	10531	74F00 23				DBASE	2181
				74F10	F15 RADAR RCVR 74F00 USED F111F 73VAC				DBASE	2181
				74F00	AS COMP G SELECT IS 1/3 OF F111F DATA				DBASE	2182
				74F00	F15 USED E SELECT TIME ESTIMATED BY				DBASE	2183
				74F00	TS BOUCHARD				DBASE	2184
				74F00	F15 RADAR RCVR				DBASE	2185
IG4F11	N7-F11	I		74F00 21	10	29L	232692 1060	105	DBASE	2186
IG4F11	N7-F11	IG4F01	G	74F00*23					DBASE	2187
IG4F11	N7-F11	IG4F02	G	74F00 23					DBASE	2188
IG4F12	K7-F12	E	72	74F00 23	59	29L	132691 1PTS31 1TSM		DBASE	2189
IG4F12	K7-F12	E	14	74F00 23	30	29L	132691 1PTS31 1TSM		DBASE	2190
IG4F12	K7-F12	E	14	74F00 23	30	29L	132691 1PTS31 1TSM		DBASE	2191
SG4F10	L7-F10	IG4F01	G	10531	74F00 23				DBASE	2192
				74F10	F15 RD FREQ OSC 74F10 USED F111F				DBASE	2193
				74F10	73VAC AS COMP G SELECT IS 1/3 OF				DBASE	2194
				74F00	F111F DATA				DBASE	2195
				74F00	AND VERIFIED BY TS BOUCHARD				DBASE	2196
				74F00	F15 RD FREQ OSC				DBASE	2197
IG4F11	N7-F11	I		74F00 21	10	29L	232692 1060	105	DBASE	2198
IG4F11	N7-F11	IG4F01	G	74F00*23					DBASE	2199
IG4F11	N7-F11	IG4F02	G	74F00 23					DBASE	2200
IG4F12	K7-F12	E	72	74F00 23	55	29L	132691 1PTS31 1TSM		DBASE	2201
IG4F12	K7-F12	E	13	74F00 23	40	29L	132691 1PTS31 1TSM		DBASE	2202
IG4F12	K7-F12	E	13	74F00 23	40	29L	132691 1PTS31 1TSM		DBASE	2203
				74F00	F15 ANT 74F00 F111F 73VAC DATA USED				DBASE	2204
				74F00	VERIFIED BY TS BOUCHARD F111 TIME USED				DBASE	2205
				74F00	F15 ANT				DBASE	2206
SG4F10	L7-F10	IG4F01	G	10531	74F00 23				DBASE	2207
IG4F11	N7-F11	E	67	74F00*23	60	29L	232691		DBASE	2208
IG4F11	K7-F11	E	16	74F00*23	30	29L	232691		DBASE	2209
IG4F11	N7-F11	E	17	74F00*23	46	29L	232691		DBASE	2210
IG4F11	N7-F11	E		74F00 21	35	29L	232692 1060	105	DBASE	2211
SG4F10	L7-F10	IG4F01	G	10531	74F00 23				DBASE	2212
				74F00	F15 RADAR TGT PROCESSOR 74F00 F111F				DBASE	2213
				74F00	73VAC DATA USED WORK AROUND				DBASE	2214
				74F00	VERIFIED BY TS BOUCHARD				DBASE	2215
				74F00	F15 RADAR TGT PROCESSOR				DBASE	2216
IG4F11	N7-F11	E		74F00 21	10	29L	232692 1060	105	DBASE	2217
IG4F11	N7-F11	IG4F01	G	74F00*23					DBASE	2218
IG4F11	N7-F11	IG4F02	G	74F00 23					DBASE	2219
IG4F12	K7-F12	E	59	74F00 23	81	29L	132691 1PTS43 1TSM		DBASE	2220
IG4F12	K7-F12	E	24	74F00 23	45	29L	132691 1PTS43 1TSM		DBASE	2221
IG4F12	K7-F12	E	17	74F00 23	45	29L	132691 1PTS43 1TSM		DBASE	2222
SG4F10	L7-F10	IG4F01	G	10531	74F00 23				DBASE	2223
				74F00	VERIFIED BY WORK AROUND				DBASE	2224
				74F00	F15 DATA PROCESSOR 74F00 F15 DATA				DBASE	2225
IG4F11	N7-F11	I		74F00 21	10	29L	232692 1060	105	DBASE	2226
IG4F11	N7-F11	IG4F01	G	74F00*23					DBASE	2227
IG4F11	N7-F11	IG4F02	G	74F00 23					DBASE	2228
IG4F12	K7-F12	E	72	74F00 23	69	29L	132691 1PTS32 1TSM		DBASE	2229
IG4F12	K7-F12	E	13	74F00 23	40	29L	132691 1PTS32 1TSM		DBASE	2230
IG4F12	K7-F12	E	13	74F00 23	40	29L	132691 1PTS32 1TSM		DBASE	2231
SG4F10	L7-F10	IG4F01	G	10531	74F00 23				DBASE	2232
				74F00	VERIFIED BY WORK AROUND				DBASE	2233
				74F00	F15 ANALOG PROCESSOR 74F00 F15 DATA				DBASE	2234
IG4F11	N7-F11	I		74F00 21	10	29L	232692 1060	105	DBASE	2235
IG4F11	N7-F11	IG4F01	G	74F00*23					DBASE	2236
IG4F11	N7-F11	IG4F02	G	74F00 23					DBASE	2237
IG4F12	K7-F12	E	63	74F00 23	90	29L	132691 1PTS33 1TSM		DBASE	2238
IG4F12	K7-F12	E	14	74F00 23	64	29L	132691 1PTS33 1TSM		DBASE	2239
IG4F12	N7-F12	E	43	74F00 23	64	29L	132691 1PTS33 1TSM		DBASE	2240
SG4F10	L7-F10	IG4F01	G	10531	74F00 23				DBASE	2241

	4		74FJ0	F15 RACAR SET CONT 74FK7 F111F	DBASE	2242
	M		74FJ0	VERIFIED BY TS OFFLEY	DBASE	2243
	4		74FJ0	F15 RACAR SET CONT	DBASE	2244
IG4FK1	47-FK1	E	11 74FJ0*23	27 29L 132631	DBASE	2245
IG4FK1	H7-FK1	E	90 74FJ0*23	26 29L 132631	DBASE	2246
IG4FK1	07-FK1	I	74FJ0 21	13 29L 232642 1060 105	DBASE	2247
SG4F1	L7-FH1	IG4FH1	G .1786 74FJ0 23		DBASE	2248
	4		74FJ0	F15 PWF SPLY 745MT F15 DATA G SELECT	DBASE	2249
	M		74FJ0	TIMES AND E SELECT GIVEN BY TSBOUCHARD	DBASE	2250
	4		74FJ0	F15 PWF SPLY 745MT	DBASE	2251
IG4F1	H7-FH1	E	57 74FJ0*23	30 29L 132631	DBASE	2252
IG4F1	K7-FH1	E	24 74FJ0*23	00 29L 132631	DBASE	2253
IG4F1	H7-FH1	E	10 74FJ0*23	12 29L 132631	DBASE	2254
IG4F1	07-FH1	I	74FJ0 21	13 29L 232642 1060 105	DBASE	2255
	4		74J00	F15 INDICTOP GROUP A70 73AEC+73EB	DBASE	2256
	4		74J00	USED F15 453MA 34	DBASE	2257
	M		74J00	VERIFIED BY TSGT FAY	DBASE	2258
	4		74J00	F15 INDICTOP GROUP	DBASE	2259
G4J1	F7-J1	G4J1	F 35 74J00 21		DBASE	2260
G4J1	07-J1	G4J1	D 74J00 21		DBASE	2261
G4J1	07-J1	G4J1	D 74J00 21		DBASE	2262
G4J1	07-J1	G4J1	E 47 74J00 23	21 29L 232642 1060 105	DBASE	2263
G4J1	47-J1	G4J1	E 17 74J00 21	12 29L 232642 1060 105	DBASE	2264
G4J1	07-J1	SG4J1	D 74J00 23		DBASE	2265
SG4J1	L7-J1	IG4J1	G .2263 74J00 23		DBASE	2266
	4		74J00	F15 MULT IND A70 73AEC DATA USED	DBASE	2267
	M		74J00	VERIFIED BY TS BOUCHARD	DBASE	2268
	4		74J00	F15 MULT IND	DBASE	2269
IG4J1	07-J1	IG4J1	I 74J00 21	20 29L 232642 1060 105	DBASE	2270
IG4J1	07-J1	IG4J1	D 74J00 23		DBASE	2271
IG4J1	07-J1	IG4J1	D 74J00 23		DBASE	2272
IG4J1	H7-J1	IG4J1	E 8 74J00 23	43 29L 132641 1PTS46 1TSD	DBASE	2273
IG4J1	K7-J1	IG4J1	E 16 74J00 23	15 29L 132641 1PTS46 1TSD	DBASE	2274
IG4J1	H7-J1	IG4J1	E 7 74J00 23	15 29L 132641 1PTS46 1TSD	DBASE	2275
SG4J1	L7-J1	IG4J1	G .2291 74J00 23		DBASE	2276
	4		74J00	F15 SIG DATA PROCESSOR 74J00 A70 73EB	DBASE	2277
	M		74J00	DATA USED WORK AROUND	DBASE	2278
	4		74J00	VERIFIED BY TS BOUCHARD	DBASE	2279
	M		74J00	F15 SIG DATA PROCESSOR	DBASE	2280
IG4J1	07-J1	IG4J1	I 74J00 21	8 29L 232642 1060 105	DBASE	2281
IG4J1	07-J1	IG4J1	D 74J00*23		DBASE	2282
IG4J1	07-J1	IG4J1	D 74J00 23		DBASE	2283
IG4J1	H7-J1	IG4J1	E 57 74J00 23	66 29L 132641 1PTS37 1TSD	DBASE	2284
IG4J1	K7-J1	IG4J1	E 21 74J00 23	49 29L 132641 1PTS37 1TSD	DBASE	2285
IG4J1	H7-J1	IG4J1	E 23 74J00 23	49 29L 132641 1PTS37 1TSD	DBASE	2286
	4		74K00	F15 HUD NETWOPK A70 73AEC+73EB DATA	DBASE	2287
	4		74K00	USED F15 453MA 16 A70 13 USE F15	DBASE	2288
	M		74K00	VERIFIED BY TSGT RAY	DBASE	2289
	4		74K00	F15 HUD	DBASE	2290
G4K1	F7-K1	G4K1	F 14 74K00 21		DBASE	2291
G4K1	07-K1	G4K1	D 74K00 21		DBASE	2292
G4K1	07-K1	G4K1	D 74K00 21		DBASE	2293
G4K1	H7-K1	G4K1	E 8 74K00 21	23 29L 232642 1060 105	DBASE	2294
G4K1	H7-K1	G4K1	E 21 74K00 21	10 29L 232642 1060 105	DBASE	2295
G4K1	07-K1	SG4K1	D 74K00 23		DBASE	2296
SG4K1	L7-K1	IG4K1	G .2291 74K00 23		DBASE	2297
	4		74K00	F15 SIG DATA PROCESSOR 74K00 A70 73EB	DBASE	2298
	M		74K00	DATA USED WORK AROUND	DBASE	2299
	4		74K00	VERIFIED BY TS BOUCHARD	DBASE	2300
	M		74K00	F15 SIG DATA PROCESSOR	DBASE	2301
IG4K1	07-K1	IG4K1	I 74K00 21	8 29L 232642 1060 105	DBASE	2302
IG4K1	07-K1	IG4K1	D 74K00*23		DBASE	2303
IG4K1	07-K1	IG4K1	D 74K00 23		DBASE	2304
IG4K1	H7-K1	IG4K1	E 57 74K00 23	66 29L 132641 1PTS36 1TSD	DBASE	2305
IG4K1	K7-K1	IG4K1	E 20 74K00 23	40 29L 132641 1PTS36 1TSD	DBASE	2306
IG4K1	H7-K1	IG4K1	E 23 74K00 23	20 29L 132681	DBASE	2307

SG4K3: L7-KA3 IG-KA3 G	13357	74K00	23						OBASE 2333
	4	74K00		F15 HUD DISP UNIT 7-KA3 A70 73504 DAT	30				OBASE 2332
	4	74K00		USED				WORK ARGUNC	OBASE 2331
	4	74K00		VERIFIED BY TS BOUCHARD					OBASE 2330
	4	74K00		F15 HUD DISP UNIT					OBASE 2329
IG4K3: 07-KA3	1	74K00	21	05 29L 1326A1 1PIS41 1TSD					OBASE 2328
IG4K3: 07-KA3 IG-KA3 J		74K00	23						OBASE 2327
IG4K3: 07-KA3 IG-KA3 E		74K00	23						OBASE 2326
IG4K2: 07-KA2	E	63	74K00	23	53 29L 1326A1 1PIS41 1TSD				OBASE 2325
IG4K2: 07-KA2	E	14	74K00	23	23 29L 1326A1 1PIS41 1TSD				OBASE 2324
IG4K2: 07-KA2	E	10	74K00	23	25 29L 1326A1 1PIS41 1TSD				OBASE 2323
SG4K3: L7-KA3 IG-KA3	1476	74K00	23						OBASE 2322
	4	74K00		VERIFIED BY TS OBERLEY					OBASE 2321
	4	74K00		F15 CAMERA 7-KA3 F15 DATA USED					OBASE 2320
IG4KE: 07-KE3	E	71	74K00	23	01 29L 1326A1				OBASE 2319
IG4KE: 07-KE3	E	25	74K00	23	25 29L 1326A1				OBASE 2318
IG4KE: 07-KE3	E	5	74K00	23	23 29L 1326A1				OBASE 2317
IG4KE: 07-KE3	E	15	74K00	21	15 29L 1326A1 1PIS41 1TSD				OBASE 2316
	4	75F00		VERIFIED BY MS GIBELYOU 2937/2164					OBASE 2315
	4	75F00		F15 PAU:2 BOMB PACK A70 75AC0 DATA					OBASE 2314
G5F03: F75F03 G5F01 E	107	75F00	21						OBASE 2313
G5F03: F75F03 G5F01 J		75F00	21						OBASE 2312
G5F03: F75F03 G5F01 E	31	75F00	21	15 29L 2462Y0 1060 105					OBASE 2311
G5F03: F75F03 G5F01 E	69	75F00	21	11 29L 2462Y0 1060 105					OBASE 2310
G5F03: F75F03 G5F01 J		75F00	23						OBASE 2309
SG5F03: L75FA3 IG5FA3		75F00	23						OBASE 2308
	4	75F00		VERIFIED BY MS GIBELYOU					OBASE 2307
	4	75F00		F15 PAU:2 A70 DAT USED					OBASE 2306
IG5FA3: 075FA3	E	87	75F00	23	35 29L 2462Y0				OBASE 2305
IG5FA3: 075FA3	E	1	75F00	23	6 29L 2462Y0				OBASE 2304
IG5FA3: 075FA3	E	10	75F00	23	11 29L 2462Y0				OBASE 2303
IG5FA3: 075FA3	E	13	75F00	21	13 29L 2462Y0 1060 105				OBASE 2302
	4	75000		VERIFIED BY MS GIBELYOU					OBASE 2301
	4	75000		F15 INSBOARD PYLON 75DA3 A70 758P0 DATA					OBASE 2300
G5003: F75003 G5001 E	152	75000	21						OBASE 2299
G5003: F75003 G5001 J		75000	21						OBASE 2298
G5003: F75003 G5001 E	1	75000	21	9 29L 3462Y0 1424X0 1060 105					OBASE 2297
G5003: F75003 G5001 E	29	75000	21	21 29L 1531Y3					OBASE 2296
G5003: F75003 G5001 E	57	75000	21	14 29L 1531Y3					OBASE 2295
	4	75000		VERIFIED BY MS GIBELYOU					OBASE 2294
	4	75000		F15 ARM LAUNCHERS NETWORK F42 AND A70					OBASE 2293
G5003: F75003 G5001 E	51	75000	21						OBASE 2292
G5003: F75003 G5001 J		75000	21						OBASE 2291
G5003: F75003 G5001 E	29	75000	21	24 29L 3462Y0 1060 105					OBASE 2290
G5003: F75003 G5001 E	21	75000	21	10 29L 2462Y0 1060					OBASE 2289
G5003: F75003 G5001 E	21	75000	21	10 29L 1531Y3 1060					OBASE 2288
G5003: F75003 G5001 E	29	75000	21	18 29L 1531Y3					OBASE 2287
G5003: F75003 G5001 E	7	75000	21	20 29L 2431Y1					OBASE 2286
SG5003: L75003 IG5003	1216	75000	23						OBASE 2285
	4	75000		VERIFIED BY MS GIBELYOU					OBASE 2284
	4	75000		F15 AIM7 LAU F42 DATA USED					OBASE 2283
IG5003: 075003	E	86	75000	23	15 29L 2462Y0				OBASE 2282
IG5003: 075003	E	9	75000	23	9 29L 1531Y3				OBASE 2281
IG5003: 075003	E	5	75000	23	10 29L 2462Y0				OBASE 2280
IG5003: 075003	E	25	75000	21	25 29L 3462Y0 1060 105				OBASE 2279
SG5003: L75003 IG5003	10520	75000	23						OBASE 2278
	4	75000		VERIFIED BY MS GIBELYOU					OBASE 2277
	4	75000		F15 AIM9 LAU A70 75AKC DATA USED					OBASE 2276
IG5003: 075003	E	95	75000	23	15 29L 2462Y0				OBASE 2275
IG5003: 075003	E	2	75000	23	3 29L 1531Y3				OBASE 2274
IG5003: 075003	E	3	75000	23	15 29L 2462Y0				OBASE 2273
IG5003: 075003	E	20	75000	21	20 29L 3462Y0 1060 105				OBASE 2272

	M		75C0J	VERIFIED BY MS GIBELYOU 2937/2454	DBASE 2374
	M		75C0J	F15 MAP NETWORK A7D 75C0J DATA USED	DBASE 2375
G5C1	F75C0J	G5C01	F	135 75C0J 21	DBASE 2376
G5C2	075C07	G5C04	D	75C0J 21	DBASE 2377
G5C3	0ALGPJ	G5C02	C	75C0J 21	DBASE 2378
G5C3	475C0J			75C0J 21	DBASE 2379
				15 29L 3462YC 1060 105	DBASE 2380
				PANEL WILL FAIL 10X AS OFTEN AS A7D	DBASE 2381
				VERIFIED BY MS GIBELYOU	DBASE 2382
				F15 M59A AUG 75 IS 210	DBASE 2383
				F15 AFM CONT SET AMS 20 A7D 74C0J USED	DBASE 2384
G5M1	F75M1J	G5M01	F	213 75M0J 21	DBASE 2385
G5M2	0075M07	G5M04	D	75M0J 21	DBASE 2386
G5M3	0ALGPJ	G5M02	C	75M0J 21	DBASE 2387
G5M3	475M1J	G5M03	E	98 75M0J 21	DBASE 2388
G5M3	475M1J			25 29L 2462YJ 1060 105	DBASE 2389
G5M3	475M1J			10 29L 2462YJ 1060 105	DBASE 2390
G5M3	540P	G5M00	D	75M0J 23	DBASE 2391
G5M3	L75M1J	G5M01	S	19107 75M0J 23	DBASE 2392
				ESTIMATES BY WORK AROUND	DBASE 2393
				F15 AFM CONT PANEL 75M0J	DBASE 2394
IG5M4	075M1J	IG5M41	D	75M0J 23	DBASE 2395
IG5M4	075M1J	IG5M42	D	75M0J 23	DBASE 2396
IG5M4	475M1J			20 29L 1326A1 1PTS45 1TS0	DBASE 2397
IG5M4	475M1J			13 29L 1326A1 1PTS45 1TS0	DBASE 2398
IG5M4	475M1J			13 29L 1326A1 1PTS45 1TS0	DBASE 2399
IG5M4	L75M1J	IG5M01	S	19107 75M0J 23	DBASE 2400
				ESTIMATES BY WORK AROUND	DBASE 2401
				F15 CONVERTER PROGRAMMER 75M0J	DBASE 2402
IG5M0	075M0J	IG5M01	D	75M0J 23	DBASE 2403
IG5M0	075M0J	IG5M02	D	75M0J 23	DBASE 2404
IG5M0	075M0J	IG5M03	D	75M0J 23	DBASE 2405
IG5M0	475M0J			50 29L 1326A1 1PTS45 1TS0	DBASE 2406
IG5M0	475M0J			40 29L 1326A1 1PTS45 1TS0	DBASE 2407
IG5M0	475M0J			40 29L 1326A1 1PTS45 1TS0	DBASE 2408
				VERIFIED BY JURSINK ONLY ONCE A7D 10	DBASE 2409
				F15 DATA=56 MCATA=14 170417 F111F17	DBASE 2410
				F15 M61A1 GUN NETWORK 75M0J	DBASE 2411
G5M1	F75M1J	G5M01	F	20000 75M0J 21	DBASE 2412
G5M1	0075M07	G5M04	D	75M0J 21	DBASE 2413
G5M1	0ALGPJ	G5M02	C	75M0J 21	DBASE 2414
G5M2	475M1J	G5M03	E	01 75M0J 21	DBASE 2415
G5M2	475M1J			40 29L 2421X2 1060 1TU225	DBASE 2416
G5M2	475M1J			14 29L 2421X1 1531Y3	DBASE 2417
G5M2	475M1J			20 29L 2421X2 1060 1TU224	DBASE 2418
G5M2	475M1J	G5M03	E	07 75M0J 21	DBASE 2419
G5M2	475M1J			30 29L 3462X0 1060 105	DBASE 2420
G5M2	475M1J			16 29L 2462X0 1060 105	DBASE 2421
G5M2	475M1J			13 29L 3462X0 1060 105	DBASE 2422
G5M3	540P	G5M00	D	75M0J 23	DBASE 2423
				F15 M51A1 GUN 75M0J	DBASE 2424
G5M1	L75M1J	IG5M01	S	1246 75M0J 23	DBASE 2425
IG5M1	475M0J			06 75M0J 23	DBASE 2426
IG5M1	475M0J			30 29L 2462Y0	DBASE 2427
IG5M1	475M0J			26 29L 2462X0	DBASE 2428
IG5M1	475M0J			30 29L 2462Y0	DBASE 2429
IG5M1	475M0J			60 29L 3462X0 1060 105	DBASE 2430
G5M1	L75M1J	IG5M01	S	10307 75M0J 23	DBASE 2431
				F15 CPU ASSY 75M0J	DBASE 2432
IG5M0	475M0J			77 75M0J 23	DBASE 2433
IG5M0	475M0J			39 29L 2462X0	DBASE 2434
IG5M0	475M0J			23 29L 2462X0	DBASE 2435
IG5M0	475M0J			60 29L 3462X0 1060 105	DBASE 2436
G5M1	L75M1J	IG5M01	S	14167 75M0J 23	DBASE 2437
				F15 CONVEYOR SYS 75M0J	DBASE 2438
IG5M1	475M0J			30 29L 2462X0	DBASE 2439
IG5M1	475M0J			120 29L 3462Y0 1060	DBASE 2440
G5M1	L75M1J	IG5M01	S	10192 75M0J 23	DBASE 2441
				F15 HYD COMP GUN 75M1J	DBASE 2442
IG5M1	475M0J			75M0J 23	DBASE 2443
IG5M1	475M0J			13 29L 2421X2	DBASE 2444
IG5M1	475M0J			30 29L 2421X2	DBASE 2445

		M	76A35	F15 ALR 56 NETWORK			DBASE 2441
G6A01	F75A01	G6A01	E	12	75A01	21	DBASE 2441
G6A01	J09M07	G6A04	J		76A30	21	DBASE 2442
G6A11	JALG01	G6A02	J		75A11	21	DBASE 2443
G6A02	F75A01		E	75	76A10	21	DBASE 2444
G6A02	F75A01	G6A04	E	25	76A10	21	DBASE 2445
G6A04	F75A01		E	16	76A10	21	DBASE 2446
G6A04	F75A02		E	91	76A10	21	DBASE 2447
					76G10		DBASE 2448
G6G01	F75G01	G6G01	E	25	75G10	21	DBASE 2449
G6G01	J09M07	G6G03	J		76G10	21	DBASE 2450
G6G01	JALG01	G6G02	J		76G10	21	DBASE 2451
G6G02	F75G01		E		76G10	21	DBASE 2452
					76M10		DBASE 2453
G6M01	F76M01	G6M01	E	12	76M00	21	DBASE 2454
G6M01	J09M07	G6M03	J		76M00	21	DBASE 2455
G6M01	JALG01	G6M02	J		76M00	21	DBASE 2456
G6M02	F76M01		E	55	76M00	21	DBASE 2457
G6M02	F76M01		E	1	76M00	21	DBASE 2458
G6M02	F76M02		E	2	76M00	21	DBASE 2459
G6M02	F76M03		E	1	76M00	21	DBASE 2460
					06000		DBASE 2461
JALG01	A06000	A06000	J		06100	21	DBASE 2462
A06001	A06000	A06000	E		06000	21	DBASE 2463
A06001	A06000		E	61	06000	21	DBASE 2464
A06001	A06000		E	47	06100	21	DBASE 2465
					TTU01		DBASE 2466
GALTU	ATTU01	ATTU01	J		TTU01	21	DBASE 2467
ATTU01	ATTU01	ATTU05	E	5	TTU01	21	DBASE 2468
ATTU05	ATTU01		E	57	TTU01	21	DBASE 2469
ATTU05	ATTU01		E	50	TTU01	21	DBASE 2470
					TESTC		DBASE 2471
G51A01	PTSNK	51A01	E	16	TESTC	23	DBASE 2472
G51A01	PTSNK		E	31	TESTC	23	DBASE 2473
51A01	LPTS1	51A02	J		TESTC	23	DBASE 2474
	LPTS1				TESTC		DBASE 2475
51A02	LPTS1	51A03	J		TESTC	23	DBASE 2476
51A02	LPTS1		J		TESTC*23		DBASE 2477
G51A01	PTSNK	51A01	E	25	TESTC	23	DBASE 2478
G51A01	PTSNK		E	64	TESTC	23	DBASE 2479
G51A01	PTSNK		E		TESTC	23	DBASE 2480
51A01	LPTS2	51A02	J		TESTC	23	DBASE 2481
	LPTS2				TESTC		DBASE 2482
51A02	LPTS2	51A03	J		TESTC	23	DBASE 2483
51A02	LPTS2		J		TESTC*23		DBASE 2484
G55A01	PTSNK	55A01	E	69	TESTC	23	DBASE 2485
G55A01	PTSNK		E	91	TESTC	23	DBASE 2486
G55A01	PTSNK		E		TESTC	23	DBASE 2487
55A01	LPTS1	55A02	J		TESTC	23	DBASE 2488
	LPTS1				TESTC		DBASE 2489
55A02	LPTS1	55A03	J		TESTC	23	DBASE 2490
55A02	LPTS1		J		TESTC*23		DBASE 2491
G52A01	PTSNK	52A01	E	21	TESTC	23	DBASE 2492
G52A01	PTSNK		E	79	TESTC	23	DBASE 2493
G52A01	PTSNK		E		TESTC	23	DBASE 2494
52A01	LPTS4	52A02	J		TESTC	23	DBASE 2495
	LPTS4				TESTC		DBASE 2496
52A02	LPTS4	52A03	J		TESTC	23	DBASE 2497
52A02	LPTS4		J		TESTC*23		DBASE 2498
G41A01	PTSNK	41A01	E	16	TESTC	23	DBASE 2499
G41A01	PTSNK		E	84	TESTC	23	DBASE 2500
G41A01	PTSNK		E		TESTC	23	DBASE 2501
41A01	LPTS5	41A02	J		TESTC	23	DBASE 2502
	LPTS5				TESTC		DBASE 2503
41A02	LPTS5	41A03	J		TESTC	23	DBASE 2504
41A02	LPTS5		J		TESTC	23	DBASE 2505

61A3J3	GPTS5	0		TESTC*23		DBASE 2536
71AK1	PTSN K 71AK1	E	34	TESTC 23		DBASE 2537
071AK	PTSN K	E	66	TESTC 23		DBASE 2538
071AK	PTSN K	E		TESTC 23		DBASE 2539
71AK2	LPTS6 71AK2	0		TESTC 23		DBASE 2540
	LPTS6	4		TESTC	LFU 71AK1 T/S	DBASE 2541
71AK3	GPTS6 71AK3	0		TESTC 23	12 29L	DBASE 2542
71AK3	GPTS6	0		TESTC*23		DBASE 2543
071FE	PTSN K 71FE1	E	16	TESTC 23		DBASE 2544
071FE	PTSN K	E	84	TESTC 23		DBASE 2545
071FE	PTSN K	E		TESTC 23		DBASE 2546
71FE2	LPTS7 71FE2	0		TESTC 23		DBASE 2547
	LPTS7	4		TESTC	LFU 71FE1 T/S	DBASE 2548
71FE3	GPTS7 71FE3	0		TESTC 23	13 29L	DBASE 2549
71FE3	GPTS7	0		TESTC*23		DBASE 2550
071E3	PTSN K 7-E301	E	37	TESTC 23		DBASE 2551
071E3	PTSN K	E	63	TESTC 23		DBASE 2552
071E3	PTSN K	E		TESTC 23		DBASE 2553
71E32	LPTS4 7-E302	0		TESTC 23		DBASE 2554
	LPTS4	4		TESTC	LFU 7-E30 T/S	DBASE 2555
71E33	GPTS4 7-E303	0		TESTC 23	13 29L	DBASE 2556
71E33	GPTS4	0		TESTC*23		DBASE 2557
071A2	PTSN K 71A201	E	39	TESTC 23		DBASE 2558
071A2	PTSN K	E	62	TESTC 23		DBASE 2559
071A2	PTSN K	E		TESTC 23		DBASE 2560
71A22	LPTS9 71A22	0		TESTC 23		DBASE 2561
	LPTS9	4		TESTC	LFU 71A20 T/S	DBASE 2562
71A23	GPTS9 71A23	0		TESTC 23	13 29L	DBASE 2563
71A23	GPTS9	0		TESTC*23		DBASE 2564
051A1	PTSN K 51A101	E	16	TESTC 23		DBASE 2565
051A1	PTSN K	E	84	TESTC 23		DBASE 2566
051A1	PTSN K	E		TESTC 23		DBASE 2567
51A12	LPTS10 51A12	0		TESTC 23		DBASE 2568
	LPTS10	4		TESTC	LFU 51A10 T/S	DBASE 2569
51A13	GPTS10 51A13	0		TESTC 23	14 29L	DBASE 2570
51A13	GPTS10	0		TESTC*23		DBASE 2571
052A3	PTSN K 52A301	E	69	TESTC 23		DBASE 2572
052A3	PTSN K	E	91	TESTC 23		DBASE 2573
052A3	PTSN K	E		TESTC 23		DBASE 2574
52A32	LPTS11 52A32	0		TESTC 23		DBASE 2575
	LPTS11	4		TESTC	LFU 52A30 T/S	DBASE 2576
52A33	GPTS11 52A33	0		TESTC 23	12 29L	DBASE 2577
52A33	GPTS11	0		TESTC*23		DBASE 2578
052A4	PTSN K 52A401	E	69	TESTC 23		DBASE 2579
052A4	PTSN K	E	91	TESTC 23		DBASE 2580
052A4	PTSN K	E		TESTC 23		DBASE 2581
52A42	LPTS12 52A42	0		TESTC 23		DBASE 2582
	LPTS12	4		TESTC	LFU 52A40 T/S	DBASE 2583
52A43	GPTS12 52A43	0		TESTC 23	12 29L	DBASE 2584
52A43	GPTS12	0		TESTC*23		DBASE 2585
049CA1	PTSN K 49CAA1	E	20	TESTC 23		DBASE 2586
049CA1	PTSN K	E	80	TESTC 23		DBASE 2587
049CA1	PTSN K	E		TESTC 23		DBASE 2588
49CA2	LPTS13 49CA2	0		TESTC 23		DBASE 2589
	LPTS13	4		TESTC	LFU 49CAA T/S	DBASE 2590
49CA3	GPTS13 49CA3	0		TESTC 23	13 29L	DBASE 2591
49CA3	GPTS13	0		TESTC*23		DBASE 2592
071FA	PTSN K 71FA1	E	47	TESTC 23		DBASE 2593
071FA	PTSN K	E	60	TESTC 23		DBASE 2594
071FA	PTSN K	E		TESTC 23		DBASE 2595
71FA2	LPTS14 71FA2	0		TESTC 23		DBASE 2596
	LPTS14	4		TESTC	LFU 71FA0 T/S	DBASE 2597
71FA3	GPTS14 71FA3	0		TESTC 23	13 29L	DBASE 2598
71FA3	GPTS14	0		TESTC*23		DBASE 2599
011FC	PTSN K 11FC1	E	37	TESTC 23		DBASE 2570
011FC	PTSN K	E	63	TESTC 23		DBASE 2571

C11P00	CTSC	0	TESTC 23	DBASE 2572
11P011	LPTS15	3	TESTC 23	DBASE 2573
	LPTS15	4	TESTC LRU 11P01 T/S	DBASE 2574
11P012	QPTS15	0	TESTC 23	DBASE 2575
11P013	QPTS15	0	TESTC*23	DBASE 2576
C55C01	PTSN.K	E	17 TESTC 23	DBASE 2577
C55C03	PTSN.K	E	82 TESTC 23	DBASE 2578
C55C03	PTSC	0	TESTC 23	DBASE 2579
55C011	LPTS15	3	TESTC 23	DBASE 2580
	LPTS15	4	TESTC LRU 55C01 T/S	DBASE 2581
55C012	QPTS15	0	TESTC 23	DBASE 2582
55C013	QPTS15	0	TESTC*23	DBASE 2583
C52A01	PTSN.K	E	16 TESTC 23	DBASE 2584
C52A01	PTSN.K	E	84 TESTC 23	DBASE 2585
C52A01	PTSC	0	TESTC 23	DBASE 2586
52A011	LPTS17	0	TESTC 23	DBASE 2587
	LPTS17	4	TESTC LRU 52A01 T/S	DBASE 2588
52A012	QPTS17	0	TESTC 23	DBASE 2589
52A013	QPTS17	0	TESTC*23	DBASE 2590
C71F01	PTSN.K	E	89 TESTC 23	DBASE 2591
C71F01	PTSN.K	E	91 TESTC 23	DBASE 2592
C71F01	PTSC	0	TESTC 23	DBASE 2593
71F011	LPTS19	0	TESTC 23	DBASE 2594
	LPTS19	4	TESTC LRU 71F01 T/S	DBASE 2595
71F012	QPTS19	0	TESTC 23	DBASE 2596
71F013	QPTS19	0	TESTC*23	DBASE 2597
C55C01	PTSN.K	E	19 TESTC 23	DBASE 2598
C55C01	PTSN.K	E	85 TESTC 23	DBASE 2599
C55C01	PTSC	0	TESTC 23	DBASE 2600
55C011	LPTS19	0	TESTC 23	DBASE 2601
	LPTS19	4	TESTC LRU 55C01 T/S	DBASE 2602
55C012	QPTS19	0	TESTC 23	DBASE 2603
55C013	QPTS19	0	TESTC*23	DBASE 2604
C57A01	PTSN.K	E	36 TESTC 23	DBASE 2605
C57A01	PTSN.K	E	64 TESTC 23	DBASE 2606
C57A01	PTSC	0	TESTC 23	DBASE 2607
57A011	LPTS22	0	TESTC 23	DBASE 2608
	LPTS22	4	TESTC LRU 57A01 T/S	DBASE 2609
57A012	QPTS22	0	TESTC 23	DBASE 2610
57A013	QPTS22	0	TESTC*23	DBASE 2611
C41A01	PTSN.K	E	27 TESTC 23	DBASE 2612
C41A01	PTSN.K	E	81 TESTC 23	DBASE 2613
C41A01	PTSC	0	TESTC 23	DBASE 2614
41A011	LPTS22	0	TESTC 23	DBASE 2615
	LPTS22	4	TESTC LRU 41A01 T/S	DBASE 2616
41A012	QPTS22	0	TESTC 23	DBASE 2617
41A013	QPTS22	0	TESTC*23	DBASE 2618
C41A01	PTSN.K	E	29 TESTC 23	DBASE 2619
C41A01	PTSN.K	E	82 TESTC 23	DBASE 2620
C41A01	PTSC	0	TESTC 23	DBASE 2621
41A011	LPTS23	0	TESTC 23	DBASE 2622
	LPTS23	4	TESTC LRU 41A01 T/S	DBASE 2623
41A012	QPTS23	0	TESTC 23	DBASE 2624
41A013	QPTS23	0	TESTC*23	DBASE 2625
C51A01	PTSN.K	E	22 TESTC 23	DBASE 2626
C51A01	PTSN.K	E	73 TESTC 23	DBASE 2627
C51A01	PTSC	0	TESTC 23	DBASE 2628
51A011	LPTS24	0	TESTC 23	DBASE 2629
	LPTS24	4	TESTC LRU 51A01 T/S	DBASE 2630
51A012	QPTS24	0	TESTC 23	DBASE 2631
51A013	QPTS24	0	TESTC*23	DBASE 2632
C51E01	PTSN.K	E	33 TESTC 23	DBASE 2633
C51E01	PTSN.K	E	67 TESTC 23	DBASE 2634
C51E01	PTSC	0	TESTC 23	DBASE 2635
51E011	LPTS25	0	TESTC 23	DBASE 2636
	LPTS25	4	TESTC LRU 51E01 T/S	DBASE 2637

51E402	QPTS25	51E403	J		TESTC 23	12 29L		DBASE 2639
51E412	QPTS25		J		TESTC*23			DBASE 2639
C5124	PTSN K	51AM01	E	16	TESTC 23			DBASE 2640
C5124	PTSN K		E	84	TESTC 23			DBASE 2641
C5124	CTSC		C		TESTC 23			DBASE 2642
51AM02	LPTS25	51AM02	J		TESTC 23			DBASE 2643
	LPTS26		4		TESTC	LPU 51AM0 T/S		DBASE 2644
51AM02	QPTS25	51AM02	J		TESTC 23	14 29L		DBASE 2645
51AM03	QPTS25		J		TESTC*23			DBASE 2646
C5221	PTSN K	52AL01	E	17	TESTC 23			DBASE 2647
C5221	PTSN K		E	83	TESTC 23			DBASE 2648
C5221	CTSC		C		TESTC 23			DBASE 2649
52AL02	LPTS27	52AL02	J		TESTC 23			DBASE 2650
52AL02	QPTS27	52AL03	J		TESTC 23			DBASE 2651
52AL03	QPTS27		J		TESTC 23			DBASE 2652
			4		TESTM	T/S FOR MICROWAVE EQUIP		DBASE 2653
C74FJ0	PTSN K	74FJ01	E	36	TESTM 23			DBASE 2654
C74FJ0	PTSN K		E	64	TESTM 23			DBASE 2655
C74FJ0	CTSM		C		TESTM 23			DBASE 2656
74FJ01	LPTS30	74FJ02	J		TESTM 23			DBASE 2657
	LPTS31		4		TESTM	LRU 74FJ0 T/S		DBASE 2658
74FJ02	QPTS30	74FJ03	J		TESTM 23	15 29L		DBASE 2659
74FJ03	QPTS30		J		TESTM*23			DBASE 2660
C74FC0	PTSN K	74FC01	E	36	TESTM 23			DBASE 2661
C74FC0	PTSN K		E	64	TESTM 23			DBASE 2662
C74FC0	CTSM		C		TESTM 23			DBASE 2663
74FC01	LPTS31	74FC02	J		TESTM 23			DBASE 2664
	LPTS31		4		TESTM	LRU 74FC0 T/S		DBASE 2665
74FC02	QPTS31	74FC03	J		TESTM 23	15 29L		DBASE 2666
74FC03	QPTS31		J		TESTM*23			DBASE 2667
C74FQ0	PTSN K	74FQ01	E	46	TESTM 23			DBASE 2668
C74FQ0	PTSN K		E	94	TESTM 23			DBASE 2669
C74FQ0	CTSM		C		TESTM 23			DBASE 2670
74FQ01	LPTS32	74FQ02	J		TESTM 23			DBASE 2671
	LPTS32		4		TESTM	LRU 74FQ0 T/S		DBASE 2672
74FQ02	QPTS32	74FQ03	J		TESTM 23	25 29L		DBASE 2673
74FQ03	QPTS32		J		TESTM*23			DBASE 2674
C74FS0	PTSN K	74FS01	E	17	TESTM 23			DBASE 2675
C74FS0	PTSN K		E	83	TESTM 23			DBASE 2676
C74FS0	CTSM		C		TESTM 23			DBASE 2677
74FS01	LPTS33	74FS02	J		TESTM 23			DBASE 2678
	LPTS33		4		TESTM	LRU 74FS0 T/S		DBASE 2679
74FS02	QPTS33	74FS03	J		TESTM 23	14 29L		DBASE 2680
74FS03	QPTS33		J		TESTM*23			DBASE 2681
C76CA0	PTSN K	76CA01	E	98	TESTM 23			DBASE 2682
C76CA0	PTSN K		E	94	TESTM 23			DBASE 2683
C76CA0	CTSM		C		TESTM 23			DBASE 2684
76CA01	LPTS34	76CA02	J		TESTM 23			DBASE 2685
	LPTS34		4		TESTM	LRU 76CA0 T/S		DBASE 2686
76CA02	QPTS34	76CA03	J		TESTM 23	25 29L		DBASE 2687
76CA03	QPTS34		J		TESTM*23			DBASE 2688
			4		TESTD	T/S FOR DISPLAYS		DBASE 2689
C13MA0	PTSN K	13MA01	E	17	TESTD 23			DBASE 2690
C13MA0	PTSN K		E	83	TESTD 23			DBASE 2691
C13MA0	CTSD		C		TESTD 23			DBASE 2692
13MA01	LPTS35	13MA02	J		TESTD 23			DBASE 2693
	LPTS35		4		TESTD	LRU 13MA0 T/S		DBASE 2694
13MA02	QPTS35	13MA03	J		TESTD 23	13 29L		DBASE 2695
13MA03	QPTS35		J		TESTD*23			DBASE 2696
C74KC0	PTSN K	74KC01	E	30	TESTD 23			DBASE 2697
C74KC0	PTSN K		E	68	TESTD 23			DBASE 2698
C74KC0	CTSD		C		TESTD 23			DBASE 2699
74KC01	LPTS36	74KC02	J		TESTD 23			DBASE 2700
	LPTS36		4		TESTD	LRU 74KC0 T/S		DBASE 2701
74KC02	QPTS36	74KC03	J		TESTD 23	12 29L		DBASE 2702
74KC03	QPTS36		J		TESTD*23			DBASE 2703

C74J00	PTSNK 7-JC01	E	19	TESTO 23		DBASE 2704
C74J01	PTSNK	E	91	TESTO 23		DBASE 2705
C74J02	PTSD	C		TESTO 23		DBASE 2706
Z4JC01	LPTS37 74JC01	J		TESTO 23		DBASE 2707
	LPTS37	4		TESTO	LRU 74JC1 T/S	DBASE 2708
Z4JC02	QPTS37 74JC02	J		TESTO 23	12 29L	DBASE 2709
74JC03	GPTS37	J		TESTO*23		DBASE 2710
C6534	PTSNK 6534	E	1	TESTO 23		DBASE 2711
C6534	PTSNK	E	91	TESTO 23		DBASE 2712
C6534	PTSD	C		TESTO 23		DBASE 2713
653411	LPTS39 653402	J		TESTO 23		DBASE 2714
	LPTS39	4		TESTO	LRU 65340 T/S	DBASE 2715
653402	QPTS39 653403	J		TESTO 23	12 29L	DBASE 2716
653403	GPTS39	J		TESTO*23		DBASE 2717
6463FA	PTSNK 6463FA1	E	16	TESTO 23		DBASE 2718
6463FA	PTSNK	E	94	TESTO 23		DBASE 2719
6463FA	PTSD	C		TESTO 23		DBASE 2720
6463FA1	LPTS39 6463FA2	J		TESTO 23		DBASE 2721
	LPTS39	4		TESTO	LRU 6463FA T/S	DBASE 2722
6463FA2	QPTS39 6463FA3	J		TESTO 23	13 29L	DBASE 2723
6463FA3	GPTS39	J		TESTO*23		DBASE 2724
51NA11	PTSNK 51NA11	E	12	TESTO 23		DBASE 2725
51NA11	PTSNK	E	63	TESTO 23		DBASE 2726
51NA11	PTSD	C		TESTO 23		DBASE 2727
51NA12	LPTS-1 51NA12	J		TESTO 23		DBASE 2728
	LPTS-1	4		TESTO	LRU 51NA1 T/S	DBASE 2729
51NA13	QPTS-1 51NA13	J		TESTO 23	12 29L	DBASE 2730
51NA13	GPTS-1	J		TESTO*23		DBASE 2731
74KA01	PTSNK 74KA01	E	24	TESTO 23		DBASE 2732
74KA01	PTSNK	E	75	TESTO 23		DBASE 2733
74KA01	PTSD	C		TESTO 23		DBASE 2734
74KA01	LPTS-1 74KA02	J		TESTO 23		DBASE 2735
	LPTS-1	4		TESTO	LRU 74KA0 T/S	DBASE 2736
74KA02	QPTS-1 74KA03	J		TESTO 23	13 29L	DBASE 2737
74KA03	GPTS-1	J		TESTO*23		DBASE 2738
51NB11	PTSNK 51NB11	E	32	TESTO 23		DBASE 2739
51NB11	PTSNK	E	68	TESTO 23		DBASE 2740
51NB11	PTSD	C		TESTO 23		DBASE 2741
51NB12	LPTS-2 51NB12	J		TESTO 23		DBASE 2742
	LPTS-2	4		TESTO	LRU 51NB1 T/S	DBASE 2743
51NB13	QPTS-2 51NB13	J		TESTO 23	12 29L	DBASE 2744
51NB13	GPTS-2	J		TESTO*23		DBASE 2745
74FF01	PTSNK 74FF01	E	19	TESTO 23		DBASE 2746
74FF01	PTSNK	E	91	TESTO 23		DBASE 2747
74FF01	PTSD	C		TESTO 23		DBASE 2748
74FF01	LPTS-3 74FF02	J		TESTO 23		DBASE 2749
	LPTS-3	4		TESTO	LRU 74FF0 T/S	DBASE 2750
74FF02	QPTS-3 74FF03	J		TESTO 23	12 29L	DBASE 2751
74FF03	GPTS-3	J		TESTO*23		DBASE 2752
75MC01	PTSNK 75MC01	E	15	TESTO 23		DBASE 2753
75MC01	PTSNK	E	91	TESTO 23		DBASE 2754
75MC01	PTSD	C		TESTO 23		DBASE 2755
75MC01	LPTS-4 75MC02	J		TESTO 23		DBASE 2756
	LPTS-4	4		TESTO	LRU 75MC0 T/S	DBASE 2757
75MC02	QPTS-4 75MC03	J		TESTO 23	12 29L	DBASE 2758
75MC03	GPTS-4	J		TESTO*23		DBASE 2759
75MA11	PTSNK 75MA11	E	19	TESTO 23		DBASE 2760
75MA11	PTSNK	E	91	TESTO 23		DBASE 2761
75MA11	PTSD	C		TESTO 23		DBASE 2762
75MA11	LPTS-5 75MA12	J		TESTO 23		DBASE 2763
	LPTS-5	4		TESTO	LRU 75MA1 T/S	DBASE 2764
75MA12	QPTS-5 75MA13	J		TESTO 23	12 29L	DBASE 2765
75MA13	GPTS-5	J		TESTO*23		DBASE 2766
74JA11	PTSNK 74JA11	E	37	TESTO 23		DBASE 2767
74JA11	PTSNK	E	63	TESTO 23		DBASE 2768
74JA11	PTSD	C		TESTO 23		DBASE 2769

74JAJ2	LOTS46	74JAJ2	0	TEST0	23		DRASE	2770
	LOTS-6		4	TEST0		LEU 74JA* T/S	DRASE	2771
74JAJ2	OTS-40	74JAJ3	3	TEST0	23	23 29L	DRASE	2772
74JAJ2	OTS-5		0	TEST0*	23		DRASE	2773
			4	TSC		TSC TEST STATION	DRASE	2774
CTSC	DU1TSC	CTSC1	0	TSC	23	01 C TSC	DRASE	2775
CTSC1	DU1TSC	TSC1	0	TSC	23		DRASE	2776
TS22	CTSC	TSC2	F	65	TSC	23	DRASE	2777
TS22	CTSC	TSC3	0	TSC	23	53 29L	DRASE	2778
TS23	CTSC		0	TSC	*23		DRASE	2779
			4	TSM		TSM TEST STATION	DRASE	2780
CTSM	DU1TSM	CTSM1	0	TSM	23	01 C TSM	DRASE	2781
CTSM1	DU1TSM	TSM1	0	TSM	23		DRASE	2782
TS41	CTSM	TSM2	F	65	TSM	23	DRASE	2783
TS42	CTSM	TSM3	0	TSM	23	50 29L	DRASE	2784
TS43	CTSM		0	TSM	*23		DRASE	2785
			4	TSD		TSD TEST STATION	DRASE	2786
CTSD	DU1TSD	CTSD1	0	TSD	23	01 C TSD	DRASE	2787
CTSD1	DU1TSD	TSD1	0	TSD	23		DRASE	2788
TS21	CTSD	TSD2	F	65	TSD	23	DRASE	2789
TS22	CTSD	TSD3	0	TSD	23	50 29L	DRASE	2790
TS23	CTSD		0	TSD	*23		DRASE	2791

APPENDIX B

F-15 TFW OPERATIONS DATA FILE

APPENDIX B

F-15 TFW OPERATIONS DATA FILE

1. The operational scenarios contained in this appendix use scheduled aircraft sortie rates of .43, .74, and 1.0. Each scenario schedules aircraft flying and maintenance activity Monday through Friday. The same schedule is repeated each week during the simulation.

2. Explanation of Aircraft Mission Names:

a. PFLTF and PFLTF are dummy missions that make a specified number of corresponding F-15 and TF-15 aircraft unavailable each day.

b. AAXX designates Air-Air Missions. The third character defines aircraft type: one is a TF-15 and two is an F-15. The fourth character defines the type of aircraft processing and number of times an aircraft flies each day: one is preflight to thruflight, two is thruflight to thruflight, three is thruflight to postflight, and four is preflight to postflight.

c. CONVXX designates conversion missions. The fifth character defines the aircraft type: one is a TF-15 and two is an F-15. The sixth character defines the type of aircraft processing and number of times an aircraft flies each day: one is preflight to thruflight, two is thruflight to postflight, three is thruflight to thruflight, and four is preflight to postflight.

d. PHASF and PHASTF are F-15 and TF-15 aircraft scheduled for Phase Inspection.

e. WASHF and WASHTF are F-15 and TF-15 aircraft scheduled for Washing and Corrosion Control Inspection.

3. Explanation of Column Headings:

- a. TIME - Daily simulation time when an aircraft begins mission processing.
- b. MISSION - Mission name.
- c. A/C TYPE - self explanatory.
- d. SCHED - number of aircraft scheduled for a given mission.
- e. MIN - minimum number of aircraft required to fly a given mission.
- f. SPARE - identifies preparation of spare aircraft for a given mission. If the spare is not used for the mission it is designated for, it then becomes a spare for the next mission.
- g. PRIORITY - LCOM peculiar code that allows an order of importance to mission scheduling.
- h. TAKEOFF - scheduled takeoff time 24 hour clock.
- i. LATENESS - time remaining after scheduled takeoff time before mission cancellation.
- j. SORTIE LENGTH - sortie length in hours and minutes.

.43 OPERATIONAL SCENARIO

DAY 1

TIME	MISSION	A/C TYPE	SCHED (MIN)	(SPARE)	PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH	
29	AA21	F15	2	2	1	2	500	2+0	1+31
114	CONV21	F15	2	2	1	2	545	2+0	1+43
130	CONV11	TF15	2	2	1	2	600	2+0	1+13
144	AA11	TF15	1	1	0	2	615	2+0	1+7
215	AA14	TF15	2	2	0	2	645	2+0	2+31
229	AA21	F15	1	1	0	2	700	2+0	1+20
329	AA14	TF15	1	1	0	2	800	2+0	0+08
349	AA24	F15	3	3	0	2	820	2+0	1+19
500	CONV26	F15	2	2	0	2	930	2+0	1+23
649	AA24	F15	1	1	0	2	1120	2+0	0+54
649	AA14	TF15	2	2	0	2	1120	2+0	1+7
730	CONV12	TF15	1	1	0	2	1200	2+0	1+23
741	WASHF	F15	1	1	0	2	800	5+0	2+5
741	WASHF	TF15	1	1	0	2	800	5+0	2+5
789	PFLTF	TF15	16	2	0	3	1200	6+0	12+0
799	PFLTF	F15	31	2	0	3	1200	6+0	12+0
1009	AA24	F15	2	2	0	2	1440	2+0	1+13
1030	AA13	TF15	1	1	0	2	1500	2+0	0+56
1059	AA23	F15	2	2	0	2	1630	2+0	1+4
1230	AA23	F15	1	1	0	2	1700	2+0	1+14
1300	CONV12	TF15	1	1	0	2	1730	2+0	1+16
1315	CONV22	F15	2	2	0	2	1745	2+0	1+21

DAY 2

TIME	MISSION	A/C TYPE	SCHED (MIN)	(SPARE)	PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH	
59	AA11	TF15	2	2	1	2	530	2+0	0+57
144	CONV21	F15	2	2	1	2	615	2+0	1+18
229	CONV24	F15	2	2	0	2	700	2+0	1+25
245	AA14	TF15	2	2	0	2	715	2+0	1+9
400	AA21	F15	1	1	0	2	730	2+0	1+54
430	AA21	F15	2	2	0	2	900	2+0	1+6
430	AA14	TF15	1	1	0	2	900	2+0	1+15
510	CONV11	TF15	1	1	1	2	940	2+0	1+13
614	CONV22	F15	2	2	0	2	1045	2+0	1+15
620	AA24	F15	2	2	0	2	1050	2+0	1+25
645	AA24	F15	2	2	0	2	1115	2+0	1+3
730	WASHF	F15	1	1	0	2	800	5+0	2+7
730	AA24	F15	2	2	0	2	1200	2+0	1+52
730	AA14	TF15	1	1	0	2	1200	2+0	1+10
741	WASHF	F15	1	1	0	2	800	5+0	2+7
900	PFLTF	TF15	16	2	0	3	1200	6+0	12+0
900	PFLTF	F15	31	2	0	3	1200	6+0	12+0
929	AA23	F15	2	2	0	2	1300	2+0	1+3
1100	CONV12	TF15	1	1	0	2	1530	2+0	1+32
1310	AA23	F15	1	1	0	2	1740	2+0	1+1
1310	AA13	TF15	2	2	0	2	1740	2+0	1+13
1600	CONV14	TF15	1	1	0	2	2030	2+0	2+20

DAY 3

TIME	MISSION	A/C TYPE	SCHED (MIN)	(SPARE)	PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH	
29	AA21	F15	2	2	1	2	500	2+0	0+50
100	CONV11	TF15	1	1	0	2	530	2+0	1+20
104	CONV21	F15	2	2	1	2	535	2+0	1+14
142	AA11	TF15	2	2	1	2	615	2+0	0+57
215	AA14	TF15	1	1	0	2	645	2+0	2+25
230	AA21	F15	1	1	0	2	700	2+0	1+20
329	AA14	TF15	1	1	0	2	800	2+0	1+0
350	AA24	F15	3	3	0	2	820	2+0	1+10
430	AA21	F15	1	1	1	2	900	2+0	0+56
499	CONV24	F15	3	3	0	2	930	2+0	1+22
557	AA22	F15	2	2	0	2	1030	2+0	1+10
647	AA24	F15	1	1	0	2	1120	2+0	1+5
690	AA14	TF15	1	1	0	2	1120	2+0	0+55
700	CONV14	TF15	1	1	0	2	1130	2+0	1+29
730	WASHF	F15	1	1	0	2	800	5+0	2+6
730	CONV12	TF15	1	1	0	2	1200	2+0	1+29
741	WASHF	F15	1	1	0	2	800	5+0	2+7
799	PFLTF	TF15	16	2	0	3	1200	6+0	12+0
799	PFLTF	F15	29	2	0	3	1200	6+0	12+0
1010	AA24	F15	1	1	0	2	1440	2+0	1+13
1030	AA13	TF15	2	2	0	2	1500	2+0	0+51
1229	AA23	F15	2	2	0	2	1700	2+0	1+5
1314	CONV22	F15	1	1	0	2	1745	2+0	1+23
1330	CONV22	F15	1	1	0	2	1800	2+0	1+13
1345	AA23	F15	2	2	0	2	1815	2+0	1+17

DAY 6

TIME	MISSION	A/C TYPE	SCHED (MIN) (SPARE)			PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH
100	AA11	TF15	1	1	1	2	930	2+ 0	0+50
149	CONV21	F15	3	3	1	2	615	2+ 0	1+23
169	AA21	F15	2	2	1	2	620	2+ 0	0+55
248	CONV24	F15	1	1	0	2	710	2+ 0	1+24
244	AA14	TF15	2	2	0	2	715	2+ 0	1+12
300	AA21	F15	3	3	0	2	730	2+ 0	2+11
329	AA14	TF15	1	1	0	2	800	2+ 0	0+59
400	CONV14	TF15	2	2	1	2	830	2+ 0	1+58
430	AA21	F15	2	2	0	2	900	2+ 0	1+ 1
430	AA14	TF15	1	1	0	2	900	2+ 0	1+ 3
644	AA24	F15	1	1	0	2	1115	2+ 0	0+55
730	WASMF	F15	1	1	0	2	800	5+ 0	2+ 5
741	PHASTF	F15	1	1	0	2	800	5+ 0	2+ 5
759	PFLTF	F15	31	2	0	3	1200	6+ 0	12+ 0
759	PFLTYF	TF15	14	2	0	3	1200	6+ 0	12+ 0
870	AA23	F15	3	3	0	2	1300	2+ 0	0+57
900	AA24	F15	1	1	0	2	1330	2+ 0	1+24
900	AA14	TF15	1	1	0	2	1330	2+ 0	1+17
919	CONV22	F15	3	3	0	2	1350	2+ 0	1+31
1245	AA21	F15	2	2	0	2	1715	2+ 0	1+19
1310	AA23	F15	1	1	0	2	1740	2+ 0	1+ 4
1310	AA13	TF15	1	1	0	2	1740	2+ 0	1+11
1330	AA23	F15	1	1	0	2	1800	2+ 0	0+58

DAY 5

TIME	MISSION	A/C TYPE	SCHED (MIN) (SPARE)			PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH
30	AA21	F15	2	2	1	2	500	2+ 0	0+59
99	CONV11	TF15	1	1	0	2	930	2+ 0	1+23
142	AA11	TF15	3	3	1	2	615	2+ 0	1+ 2
200	CONV14	TF15	1	1	0	2	630	2+ 0	1+45
229	AA21	F15	2	2	1	2	700	2+ 0	1+28
300	CONV21	F15	3	3	1	2	730	2+ 0	1+53
390	AA24	F15	3	3	0	2	820	2+ 0	1+17
500	CONV24	F15	1	1	0	2	930	2+ 0	1+16
929	AA24	F15	2	2	0	2	1000	2+ 0	1+ 1
647	AA24	F15	1	1	0	2	1120	2+ 0	1+ 3
650	AA14	TF15	1	1	0	2	1120	2+ 0	0+53
730	WASMTF	TF15	1	1	0	2	800	5+ 0	2+ 6
790	CONV12	TF15	1	1	0	2	1200	2+ 0	1+25
742	PHASTF	TF15	1	1	0	2	800	5+ 0	2+ 5
800	PFLTF	F15	30	2	0	3	1200	6+ 0	12+ 0
800	PFLTYF	TF15	15	2	0	3	1200	6+ 0	12+ 0
1010	AA24	F15	1	1	0	2	1440	2+ 0	1+ 7
1030	AA13	TF15	3	3	0	2	1500	2+ 0	0+55
1100	AA23	F15	2	2	0	2	1930	2+ 0	1+11
1230	AA23	F15	2	2	0	2	1700	2+ 0	1+23
1715	CONV22	F15	3	3	0	2	1745	2+ 0	1+58

.74 OPERATIONAL SCENARIO

DAY 1

TIME	MISSION	A/C TYPE	SCMD	THIN	SPAE	PRIORITY	TAKEOFF	LATENESS	FO-TIE LENGTH
29	4421	F15	2	2	1	2	57	2+	1+7
100	CONV11	TF15	1	1	0	2	57	2+	1+11
116	CONV21	F15	2	2	1	2	60	2+	1+12
130	CONV11	TF15	2	2	1	2	60	2+	1+15
145	4421	TF15	1	1	0	2	61	2+	1+14
159	CONV1-	TF15	4	2	1	2	63	2+	1+14
219	4421	TF15	2	2	0	2	64	2+	1+17
230	4421	F15	1	3	0	2	70	2+	1+20
329	4421	TF15	1	1	0	2	71	2+	1+7
350	4424	F15	3	3	1	2	82	2+	1+11
400	4421	F15	2	2	0	2	83	2+	1+23
414	4421	F15	2	2	0	2	84	2+	1+5
459	CONV2-	F15	2	2	0	2	93	2+	1+17
530	4424	F15	2	2	0	2	100	2+	1+53
600	4422	F15	1	2	0	2	100	2+	1+11
650	4424	F15	1	1	0	2	112	2+	1+7
690	4424	TF15	1	1	0	2	112	2+	1+2
700	CONV1-	TF15	1	1	0	2	113	2+	1+2
730	4454P	F15	1	1	0	2	90	5+	2+5
730	CONV12	TF15	1	1	0	2	120	2+	1+13
741	4454P	F15	1	1	0	2	87	5+	2+5
741	4454P	F15	1	1	0	2	87	5+	2+7
741	4454P	TF15	1	1	0	2	87	5+	2+5
759	4454P	TF15	1	2	0	3	120	6+	12+1
759	4454P	F15	18	2	0	3	120	6+	12+1
1019	4424	F15	3	3	0	2	100	2+	1+12
1030	4423	TF15	2	2	0	2	100	2+	1+7
1059	4423	F15	2	2	0	2	153	2+	1+8
1139	4423	F15	3	3	0	2	161	2+	1+1
1229	4423	F15	2	2	0	2	170	2+	1+7
1300	CONV12	TF15	1	1	0	2	170	2+	1+12
1314	CONV22	F15	2	2	0	2	174	2+	1+27
1645	4423	F15	2	2	0	2	211	2+	2+7

DAY 2

TIME	MISSION	A/C TYPE	SCMD	THIN	SPAE	PRIORITY	TAKEOFF	LATENESS	FO-TIE LENGTH
100	4421	F15	2	2	1	2	57	2+	1+5
130	CONV21	TF15	2	2	1	2	60	2+	1+17
145	CONV21	F15	2	2	1	2	61	2+	1+24
190	4421	F15	2	2	1	2	63	2+	1+14
230	CONV2-	F15	2	2	0	2	70	2+	1+12
240	4421	TF15	2	2	1	2	71	2+	1+7
300	4421	F15	3	3	1	2	73	2+	2+14
349	4421	F15	2	2	0	2	81	2+	1+17
430	4421	F15	1	1	0	2	90	2+	1+12
430	4421	TF15	1	1	0	2	90	2+	1+4
569	CONV11	TF15	2	2	0	2	94	2+	1+21
600	CONV11	TF15	2	2	0	2	100	2+	1+21
619	CONV22	F15	2	2	0	2	104	2+	1+11
629	4424	F15	2	2	0	2	109	2+	1+5
645	4424	F15	2	2	0	2	115	2+	1+13
700	4423	F15	2	2	0	2	120	2+	1+13
730	4454P	F15	1	1	0	2	90	5+	2+5
730	4424	F15	2	2	0	2	100	2+	1+11
741	4454P	TF15	1	1	0	2	87	5+	2+5
741	4454P	F15	1	1	0	2	87	5+	2+5
741	4454P	F15	1	1	0	2	87	5+	2+5
759	4454P	TF15	18	2	0	3	120	6+	12+1
759	4454P	F15	9	2	0	3	120	6+	12+1
830	4423	F15	3	3	0	2	100	2+	1+14
900	4424	F15	1	1	0	2	133	2+	1+7
900	4424	TF15	1	1	0	2	133	2+	1+14
1045	CONV22	F15	2	2	0	2	151	2+	1+14
1059	CONV12	TF15	2	2	0	2	159	2+	1+12
1130	CONV2-	F15	2	2	0	2	161	2+	1+14
1239	CONV2-	F15	1	1	0	2	171	2+	1+9
1239	CONV12	TF15	1	1	0	2	171	2+	1+12
1300	4423	F15	2	2	0	2	170	2+	1+7
1309	4423	F15	1	1	0	2	170	2+	1+1
1309	4423	TF15	1	1	0	2	170	2+	1+4
1330	4423	F15	2	2	0	2	180	2+	1+11
1600	CONV12	TF15	2	2	0	2	199	2+	1+1
1630	CONV12	TF15	2	2	0	2	211	2+	1+4

TIME	MISSION	A/D TYPE	SCHED (MIN) (SPARE)	PRIORITY	TAKEOFF	LATENESS	SOFTIE LENGTH		
30	4421	F15	2	2	1	2	530	2+	1+2
59	CONV11	TF15	1	1	0	2	530	2+	1+5
109	CONV21	F15	2	2	0	2	559	2+	1+22
130	CONV11	TF15	2	2	1	2	600	2+	1+26
142	4411	TF15	1	1	0	2	619	2+	1+1
200	CONV1-	TF15	4	2	1	2	630	2+	1+1
215	4414	TF15	2	2	0	2	649	2+	2+3
229	4421	F15	3	3	0	2	700	2+	1+15
330	4414	TF15	1	1	0	2	800	2+	0+7
349	4424	F15	3	3	0	2	820	2+	2+15
430	4421	F15	2	2	0	2	900	2+	1+1
500	CONV2-	F15	2	2	0	2	930	2+	1+20
529	4424	F15	2	2	0	2	1000	2+	1+2
557	4422	F15	2	2	0	2	1030	2+	1+20
647	4424	F15	1	1	0	2	1100	2+	1+3
649	4414	TF15	1	1	0	2	1120	2+	1+5
659	CONV1-	TF15	1	1	0	2	1130	2+	1+8
730	445MF	F15	1	1	0	2	810	5+	2+5
730	CONV2	TF15	1	1	0	2	1200	2+	1+13
734	CONV2	F15	2	2	0	2	1215	2+	1+13
741	4415F	F15	1	1	0	2	800	5+	2+8
741	4415F	F15	1	1	0	2	800	5+	2+5
800	4415F	TF15	1	1	0	2	800	5+	2+8
800	4415F	TF15	7	2	0	3	1200	6+	12+1
800	4415F	F15	20	2	0	3	1200	5+	12+0
800	4415F	TF15	1	1	0	2	1230	2+	0+3
1009	4421	F15	3	3	0	2	1000	2+	1+20
1030	4413	TF15	1	1	0	2	1500	2+	1+9
1230	4423	F15	2	2	0	2	1700	2+	1+3
1259	CONV12	TF15	2	2	0	2	1730	2+	1+22
1315	CONV22	F15	2	2	0	2	1749	2+	1+18
1330	CONV22	F15	2	2	0	2	1800	2+	1+33
1344	4423	F15	2	2	0	2	1819	2+	1+9
1604	4423	F15	2	2	0	2	2115	2+	2+24

DAY

TIME	MISSION	A/D TYPE	SCHED (MIN) (SPARE)	PRIORITY	TAKEOFF	LATENESS	SOFTIE LENGTH		
59	4421	F15	2	2	2	2	530	2+	1+2
130	CONV11	TF15	2	2	1	2	600	2+	1+34
144	CONV21	F15	2	2	2	2	619	2+	1+22
150	4421	F15	2	2	1	2	620	2+	1+3
239	CONV2-	F15	2	2	0	2	700	2+	1+19
245	4414	TF15	2	2	1	2	715	2+	1+16
300	4421	F15	3	3	0	2	730	2+	2+7
320	4414	TF15	2	2	1	2	800	2+	1+3
359	CONV1-	TF15	2	2	0	2	830	2+	2+14
430	4421	F15	1	1	0	2	900	2+	1+9
430	4424	TF15	1	1	0	2	900	2+	1+20
430	CONV11	TF15	2	2	0	2	1030	2+	1+25
649	4424	F15	2	2	0	2	1100	2+	1+2
659	4422	F15	2	2	0	2	1130	2+	1+9
709	4422	F15	2	2	0	2	1140	2+	1+2
730	445MF	TF15	1	1	0	2	800	5+	2+5
730	445MF	F15	1	1	0	2	800	5+	2+8
730	4424	F15	2	2	0	2	1200	2+	1+33
800	4415F	F15	20	2	0	3	1200	5+	12+0
800	4415F	TF15	7	2	0	3	1200	6+	12+1
829	4424	F15	2	2	0	2	1000	2+	1+8
845	CONV2-	F15	2	2	0	2	1315	2+	1+27
900	4424	F15	1	1	0	2	1330	2+	1+9
900	4414	TF15	1	1	0	2	1330	2+	1+5
920	CONV22	F15	2	2	0	2	1300	2+	1+22
1100	4424	F15	2	2	0	2	1530	2+	1+14
1240	CONV2-	F15	2	1	0	2	1710	2+	1+29
1240	CONV12	TF15	1	1	0	2	1710	2+	1+17
1249	4423	F15	2	2	0	2	1719	2+	1+10
1309	4421	F15	1	1	0	2	1740	2+	1+17
1309	4413	TF15	1	1	0	2	1740	2+	1+10
1330	4421	F15	2	2	0	2	1800	2+	1+5
1559	CONV1-	TF15	2	2	0	2	2030	2+	1+9
1630	CONV12	TF15	2	2	0	2	2100	2+	1+7

DAY 5												
TIME	MISSION	A/C	TYPE	SCHED (MIN)			SPARE	PRIORITY	TAKEOFF	LATENCY	SORTIE LENGTH	
30	AA21	P19		2	2			2	500	20	1000	
50	CONV11	TF15		1	1	C		2	530	20	1025	
130	CONV11	TF15		2	2		1	2	600	20	1027	
142	AA11	TF15		1	1	C		2	615	20	1002	
200	CONV19	TF19		6	6		1	2	630	20	1009	
215	AA14	TF15		2	2	C		2	645	20	2019	
229	AA21	P19		3	3		1	2	700	20	1019	
300	CONV21	P15		2	2		1	2	730	20	1000	
390	AA24	TF19		1	1	C		2	800	20	1000	
349	AA24	P15		3	3		1	2	820	20	1006	
399	AA20	P19		2	2			2	830	20	1009	
415	AA21	P15		2	2	C		2	845	20	1005	
500	CONV20	P15		2	2	C		2	880	20	1019	
529	AA24	P15		2	2	C		2	900	20	1052	
597	AA22	P19		2	2			2	900	20	1019	
647	AA20	P15		1	1	C		2	910	20	1003	
649	AA24	TF19		1	1	C		2	920	20	1053	
659	CONV14	TF15		1	1	C		2	930	20	1018	
730	AA24	TF19		1	1			2	950	20	1009	
730	CONV12	TF15		1	1			2	970	20	1032	
900	AA21	P19		22	22	C		3	1000	60	1203	
900	AA21	TF15		6	2	C		3	1200	60	1200	
1009	AA20	P19		3	3			2	1400	20	1001	
1030	AA13	TF15		1	1	C		2	1500	20	1056	
1100	AA21	P19		2	2	C		2	1500	20	1010	
1129	AA23	P15		3	3	C		2	1610	20	1003	
1230	AA20	P19		2	2	C		2	1700	20	1001	
1259	CONV12	TF15		1	1	C		2	1730	20	1033	
1319	CONV22	P19		2	2	C		2	1749	20	1017	
1644	AA23	P15		2	2	C		2	2115	20	1000	

1.0 OPERATIONAL SCENARIO

DAY 1									
TIME	MISSION	A/C TYPE	SCHED (MIN)	(SPARE)	PRIORITY	TAKOFF	LATENESS	SORTIE LENGTH	
29	AA21	F15	2	2	1	2	500	2+0	1+31
59	PHASF	F15	1	1	0	2	810	5+0	2+7
59	PHASF	F15	1	1	0	2	800	5+0	2+5
99	PHASTP	TF15	1	1	0	2	800	5+0	2+6
59	WASHF	F15	1	1	0	2	800	5+0	2+6
180	CONV11	TF15	1	1	0	2	530	2+0	1+28
109	AA11	TF15	2	2	1	2	540	2+0	0+53
109	AA21	F15	2	2	0	2	540	2+0	1+6
114	CONV21	F15	2	2	1	2	545	2+0	1+23
130	CONV11	TF15	2	2	1	2	600	2+0	1+23
140	AA21	F15	2	2	0	2	610	2+0	2+28
144	AA11	TF15	1	1	0	2	615	2+0	0+59
154	CONV14	TF15	4	2	1	2	630	2+0	1+31
215	AA14	TF15	2	2	0	2	645	2+0	2+6
224	AA21	F15	1	3	0	2	700	2+0	1+31
239	CONV11	TF15	1	1	0	2	710	2+0	1+21
380	CONV24	F15	2	2	1	2	730	2+0	1+15
314	CONV21	F15	1	1	0	2	745	2+0	1+26
329	AA14	TF15	1	1	0	2	800	2+0	0+54
349	AA24	F15	3	3	1	2	820	2+0	1+19
359	AA21	F15	2	2	0	2	830	2+0	1+8
415	AA21	F15	2	2	0	2	845	2+0	0+56
563	CONV24	F15	2	2	0	2	930	2+0	1+25
529	AA24	F15	2	2	0	2	1000	2+0	3+57
545	AA24	F15	1	1	0	2	1015	2+0	1+54
545	AA14	TF15	1	1	0	2	1015	2+0	2+1
680	AA22	F15	2	2	0	2	1030	2+0	1+15
649	AA24	F15	1	1	0	2	1120	2+0	0+52
649	AA14	TF15	1	1	0	2	1120	2+0	0+53
788	CONV14	TF15	1	1	0	2	1130	2+0	1+39
715	AA23	F15	2	2	0	2	1145	2+0	1+16
725	AA13	TF15	2	2	0	2	1155	2+0	1+27
730	CONV12	TF15	1	1	0	2	1200	2+0	1+32
799	PFLTF	TF15	3	2	0	3	1200	6+0	12+0
799	PFLTF	F15	11	2	0	3	1200	6+0	12+0
829	AA23	F15	2	2	0	2	1300	2+0	2+9
1009	AA24	F15	3	3	0	2	1440	2+0	1+29
1038	AA13	TF15	2	2	0	2	1500	2+0	1+3
1344	CONV22	F15	1	1	0	2	1515	2+0	1+32
1059	AA23	F15	2	2	0	2	1530	2+0	1+1
1129	CONV12	TF15	1	1	0	2	1600	2+0	1+26
1139	AA23	F15	3	3	0	2	1610	2+0	1+3
1236	AA23	F15	2	2	0	2	1700	2+0	1+0
1388	CONV12	TF15	1	1	0	2	1730	2+0	1+28
1315	CONV22	F15	2	2	0	2	1745	2+0	1+14
1845	AA23	F15	2	2	0	2	2115	2+0	2+6

DAY 2

TIME	MISSION	A/C TYPE	SCHED (MINI (SPARE))			PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH
99	AA21	F15	2	2	1	2	530	2+0	1+5
100	MASHF	F15	1	1	0	2	600	5+0	2+5
100	PHASFP	TF15	1	1	0	2	600	5+0	2+6
100	PHASF	F15	1	1	0	2	600	5+0	2+5
100	PHASF	F15	1	1	0	2	600	5+0	2+5
110	CONV24	F15	1	1	1	2	540	2+0	1+22
111	CONV24	F15	1	1	0	2	542	2+0	1+22
115	CONV24	F15	1	1	0	2	545	2+0	1+31
130	CONV11	TF15	2	2	1	2	600	2+0	1+17
144	CONV21	F15	2	2	1	2	615	2+0	1+29
150	AA21	F15	2	2	1	2	620	2+0	1+0
210	CONV11	TF15	1	1	0	2	640	2+0	1+29
229	CONV24	F15	2	2	0	2	700	2+0	1+39
245	AA14	TF15	2	2	1	2	715	2+0	1+13
300	AA21	F15	3	3	0	2	730	2+0	1+53
345	AA21	F15	2	2	0	2	815	2+0	1+16
359	AA24	F15	2	1	0	2	830	2+0	1+11
359	AA14	TF15	1	1	0	2	830	2+0	1+2
415	AA24	F15	1	1	0	2	845	2+0	1+4
415	AA14	TF15	1	1	0	2	845	2+0	0+50
430	AA21	F15	1	1	0	2	900	2+0	1+11
430	AA14	TF15	1	1	0	2	900	2+0	1+6
510	CONV11	TF15	2	2	0	2	940	2+0	1+24
529	AA21	F15	2	1	0	2	1000	2+0	1+12
529	AA11	TF15	1	1	0	2	1000	2+0	1+11
600	CONV11	TF15	2	2	0	2	1030	2+0	1+8
614	CONV22	F15	2	2	0	2	1045	2+0	1+15
620	AA24	F15	2	2	0	2	1050	2+0	1+1
645	AA24	F15	2	2	0	2	1115	2+0	0+52
659	AA23	F15	2	2	0	2	1130	2+0	0+55
730	AA24	F15	2	2	0	2	1200	2+0	1+40
800	PFLTF	TF15	5	2	0	3	1200	6+0	12+0
800	PFLTF	F15	9	2	0	3	1200	6+0	12+0
829	AA23	F15	3	3	0	2	1300	2+0	0+53
900	AA24	F15	1	1	0	2	1330	2+0	1+8
900	AA14	TF15	1	1	0	2	1330	2+0	1+24
930	CONV12	TF15	1	1	0	2	1400	2+0	1+30
1044	CONV22	F15	2	2	0	2	1515	2+0	1+31
1100	CONV12	TF15	2	2	0	2	1530	2+0	1+32
1129	CONV24	F15	2	2	0	2	1600	2+0	1+15
1240	CONV24	F15	1	1	0	2	1710	2+0	1+23
1240	CONV12	TF15	1	1	0	2	1710	2+0	1+21
1300	AA23	F15	2	2	0	2	1730	2+0	1+11
1310	AA23	F15	1	1	0	2	1740	2+0	1+11
1310	AA13	TF15	1	1	0	2	1740	2+0	1+14
1330	AA23	F15	2	2	0	2	1800	2+0	1+2
1400	AA23	F15	2	2	0	2	1830	2+0	2+18
1400	AA13	TF15	1	1	0	2	1830	2+0	2+26
1600	CONV14	TF15	2	2	0	2	2030	2+0	2+22
1630	CONV12	TF15	2	2	0	2	2100	2+0	1+16

DAY 3

TIME	MISSION	A/C TYPE	SCHED (MIN) (SPARE)			PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH
29	AA21	F15	2	2	1	2	500	2+0	1+1
59	MASHTF	TF15	1	1	0	2	800	5+0	2+6
59	PHASF	F15	1	1	0	2	800	5+0	2+5
59	PHASF	F15	1	1	0	2	800	5+0	2+6
59	PHASTF	TF15	1	1	0	2	800	5+0	2+6
59	MASHF	F15	1	1	0	2	800	5+0	2+5
100	CONV12	TF15	1	1	0	2	530	2+0	1+19
106	CONV21	F15	2	2	0	2	535	2+0	1+17
130	CONV11	TF15	2	2	1	2	600	2+0	1+23
142	AA11	TF15	1	1	0	2	615	2+0	1+9
159	CONV14	TF15	4	2	1	2	630	2+0	1+59
215	AA14	TF15	2	2	0	2	645	2+0	2+20
230	AA21	F15	3	3	0	2	700	2+0	1+23
244	AA24	F15	2	2	0	2	715	2+0	1+12
244	AA11	TF15	1	1	0	2	715	2+0	1+2
250	CONV21	F15	1	1	1	2	720	2+0	1+30
300	CONV21	F15	1	1	0	2	730	2+0	1+12
315	AA21	F15	2	2	1	2	745	2+0	2+14
315	AA11	TF15	1	1	0	2	745	2+0	2+25
329	AA14	TF15	1	1	0	2	800	2+0	1+5
356	AA24	F15	3	3	0	2	820	2+0	1+5
430	AA21	F15	3	3	0	2	900	2+0	0+50
459	CONV24	F15	2	2	0	2	930	2+0	1+35
530	AA24	F15	2	2	0	2	1000	2+0	0+54
530	CONV24	F15	1	1	0	2	1000	2+0	1+22
557	AA22	F15	2	2	0	2	1030	2+0	1+6
647	AA24	F15	1	1	0	2	1120	2+0	0+55
650	AA14	TF15	1	1	0	2	1120	2+0	0+57
700	CONV14	TF15	1	1	0	2	1130	2+0	1+15
730	CONV12	TF15	1	1	0	2	1200	2+0	1+28
735	CONV23	F15	2	2	0	2	1205	2+0	1+30
759	MPLTF	TF15	3	2	0	3	1200	6+0	12+0
759	MPLTF	F15	9	2	0	3	1200	6+0	12+0
759	AA14	TF15	1	1	0	2	1230	2+0	1+1
830	AA12	TF15	1	1	1	2	1300	2+0	1+10
830	AA24	F15	1	1	0	2	1300	2+0	1+14
1310	AA24	F15	3	3	0	2	1440	2+0	1+15
1030	AA13	TF15	1	1	0	2	1500	2+0	1+11
1130	CONV22	F15	1	1	0	2	1600	2+0	1+27
1140	CONV22	F15	1	1	0	2	1610	2+0	1+23
1200	AA24	F15	2	1	0	2	1630	2+0	0+56
1200	AA13	TF15	1	1	0	2	1630	2+0	1+2
1229	AA23	F15	2	2	0	2	1700	2+0	1+7
1300	CONV12	TF15	2	2	0	2	1730	2+0	1+25
1314	CONV22	F15	2	2	0	2	1745	2+0	1+29
1330	CONV22	F15	2	2	0	2	1800	2+0	1+36
1345	AA23	F15	2	2	0	2	1815	2+0	1+9
1430	AA23	F15	2	1	0	2	1900	2+0	0+55
1430	AA13	TF15	1	1	0	2	1900	2+0	0+52
1645	AA23	F15	2	2	0	2	2115	2+0	2+9

DAY 4

TIME	MISSION	A/C TYPE	SCHED (MIN)	(SPARE)	PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH	
59	WASHYP	TF15	1	1	0	2	800	5+0	2+5
59	WASHF	F15	1	1	0	2	800	5+0	2+6
59	PHASTF	F15	1	1	0	2	800	5+0	2+5
59	PHASTF	TF15	1	1	0	2	800	5+0	2+6
100	AA21	F15	2	2	1	2	550	2+0	1+3
130	CONV11	TF15	2	2	1	2	600	2+0	1+39
139	AA21	F15	2	2	0	2	610	2+0	2+31
139	AA11	TF15	1	1	0	2	610	2+0	2+5
149	CONV21	F15	2	2	1	2	615	2+0	1+24
149	AA21	F15	2	2	1	2	620	2+0	1+0
240	CONV24	F15	2	2	0	2	710	2+0	1+38
244	AA14	TF15	2	2	1	2	715	2+0	1+19
300	AA21	F15	3	3	0	2	750	2+0	2+9
315	CONV21	F15	1	1	0	2	745	2+0	1+36
329	AA14	TF15	2	2	1	2	800	2+0	1+12
345	AA21	F15	1	1	1	2	815	2+0	1+12
345	AA14	TF15	1	1	0	2	815	2+0	1+29
400	CONV14	TF15	2	2	0	2	830	2+0	2+21
430	AA21	F15	1	1	0	2	900	2+0	1+1
430	AA14	TF15	1	1	0	2	900	2+0	1+5
445	CONV11	TF15	1	1	0	2	915	2+0	1+31
459	CONV21	F15	1	1	0	2	930	2+0	1+36
630	CONV11	TF15	2	2	0	2	1030	2+0	1+16
645	AA24	F15	2	2	0	2	1115	2+0	0+57
700	AA22	F15	2	2	0	2	1130	2+0	1+3
710	AA22	F15	2	2	0	2	1140	2+0	1+3
730	AA24	F15	2	2	0	2	1230	2+0	1+49
759	PFLTF	F15	1	1	0	3	1200	6+0	12+0
759	PFLTF	TF15	2	2	0	3	1200	6+0	12+0
830	AA23	F15	3	3	0	2	1300	2+0	1+4
844	CONV24	F15	2	2	0	2	1315	2+0	1+33
900	AA24	F15	1	1	0	2	1330	2+0	1+13
900	AA14	TF15	1	1	0	2	1330	2+0	1+4
919	CONV22	F15	2	2	0	2	1350	2+0	1+23
929	AA13	TF15	1	1	0	2	1400	2+0	0+52
929	AA24	F15	2	1	0	2	1400	2+0	1+0
1030	AA24	F15	1	1	0	2	1500	2+0	2+6
1030	AA23	F15	1	1	0	2	1500	2+0	2+5
1059	AA24	F15	2	2	0	2	1530	2+0	1+18
1130	AA23	F15	2	2	0	2	1600	2+0	2+22
1240	CONV24	F15	1	1	0	2	1710	2+0	1+33
1240	CONV12	TF15	1	1	0	2	1710	2+0	1+23
1245	AA23	F15	2	2	0	2	1715	2+0	1+21
1310	AA23	F15	1	1	0	2	1740	2+0	1+25
1316	AA13	TF15	1	1	0	2	1740	2+0	1+5
1330	AA23	F15	2	2	0	2	1800	2+0	0+58
1500	CONV22	F15	1	1	0	2	1930	2+0	1+19
1515	CONV12	TF15	1	1	0	2	1945	2+0	1+27
1529	CONV22	F15	1	1	0	2	2000	2+0	1+28
1600	CONV14	TF15	2	2	0	2	2030	2+0	2+11
1630	CONV12	TF15	2	2	0	2	2100	2+0	1+25

DAY 5

TIME	MISSION	A/C TYPE	SCHED (MIN)	(SPARE)	PRIORITY	TAKEOFF	LATENESS	SORTIE LENGTH	
30	AA21	F15	2	2	1	2	500	2+ 0	0+54
59	CONV11	TF15	1	1	0	2	530	2+ 0	1+23
100	PHASF	F15	1	1	0	2	600	5+ 0	2+ 6
108	WASHF	F15	1	1	0	2	600	5+ 0	2+ 5
100	WASHTF	TF15	1	1	0	2	600	5+ 0	2+ 6
130	CONV11	TF15	2	2	1	2	600	2+ 0	1+39
142	AA11	TF15	1	1	0	2	615	2+ 0	0+57
200	CONV14	TF15	2	2	1	2	630	2+ 0	1+31
215	AA14	TF15	2	2	0	2	645	2+ 0	2+ 4
229	AA21	F15	3	3	1	2	700	2+ 0	1+16
300	CONV21	F15	2	2	1	2	730	2+ 0	1+59
314	AA21	F15	2	2	1	2	745	2+ 0	2+19
314	AA11	TF15	1	1	1	2	745	2+ 0	2+ 7
330	AA14	TF15	1	1	0	2	800	2+ 0	0+52
350	AA24	F15	3	3	1	2	820	2+ 0	1+10
359	AA21	F15	2	2	0	2	830	2+ 0	1+15
415	AA21	F15	2	2	0	2	845	2+ 0	1+ 6
430	AA21	F15	3	2	0	2	900	2+ 0	1+51
430	AA11	TF15	1	1	0	2	900	2+ 0	2+14
500	CONV24	F15	2	2	0	2	930	2+ 0	1+18
515	CONV21	F15	1	1	0	2	945	2+ 0	1+14
529	AA24	F15	2	2	0	2	1000	2+ 0	0+54
545	CONV11	TF15	1	1	0	2	1015	2+ 0	1+21
557	AA22	F15	2	2	0	2	1030	2+ 0	1+ 6
647	AA24	F15	1	1	0	2	1120	2+ 0	0+51
650	AA14	TF15	1	1	0	2	1120	2+ 0	0+57
659	CONV14	TF15	1	1	0	2	1130	2+ 0	1+37
730	CONV12	TF15	1	1	0	2	1200	2+ 0	1+25
800	PFLTF	F15	11	2	0	3	1200	6+ 0	12+ 0
800	PFLTF	TF15	3	2	0	3	1200	6+ 0	12+ 0
829	CONV22	F15	1	1	0	2	1300	2+ 0	1+17
865	CONV24	F15	1	1	0	2	1335	2+ 0	1+21
1010	AA24	F15	3	3	0	2	1440	2+ 0	1+ 4
1030	AA13	TF15	1	1	0	2	1500	2+ 0	1+ 3
1140	AA23	F15	3	2	0	2	1510	2+ 0	1+ 5
1140	AA14	TF15	1	1	0	2	1510	2+ 0	0+52
1100	AA23	F15	2	2	0	2	1530	2+ 0	1+11
1140	AA23	F15	3	3	0	2	1610	2+ 0	0+51
1230	AA23	F15	2	2	0	2	1730	2+ 0	1+ 9
1259	CONV12	TF15	1	1	0	2	1730	2+ 0	1+25
1315	CONV22	F15	2	2	0	2	1745	2+ 0	1+15
1344	AA23	F15	2	1	0	2	1815	2+ 0	1+23
1344	AA13	TF15	2	1	0	2	1815	2+ 0	1+21
1429	CONV12	TF15	1	1	0	2	1900	2+ 0	1+23
1500	CONV24	F15	1	1	0	2	1930	2+ 0	1+20
1644	AA23	F15	2	2	0	2	2115	2+ 0	2+37

APPENDIX C

PERFORMANCE SUMMARY REPORTS AND MATRICES

APPENDIX C

PERFORMANCE SUMMARY REPORTS AND MATRICES

Performance Summary Reports (PSR's) and on/off equipment manpower and backorder matrices are illustrated in this appendix. These statistics reflect the results of the F-15 peacetime simulation based on unconstrained parts, unconstrained avionic test stations, and a .74 scheduled sortie rate. Figures C-1, C-2, and C-3 describe the results of the unconstrained manpower simulation. Figures C-4, C-5, and C-6 contain output statistics for the constrained manpower simulation. Drake (Ref 7 and 8) describes in detail the statistical output of an LCOM simulation.

	TOTAL	6240	6310	6311	6312	6313	6320	6321	6322	6323	6324	6325	6326
17 MANHOURS AVAILABLE (17)	1270.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
18 MANHOURS AVAILABLE (18)	1270.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
19 PERCENT UTILIZATION	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00
20 MANHOURS USED (19)	1270.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
21 PCT DUES/FEES/CONTINGENT	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
22 PCT SCHED. MAINTENANCE	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
23 NUMBER OF PZ PLANNING	1270.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
24 NUMBER OF PZ AVAILABLE (20)	1270.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
25 PCT AVAILABLE (20)	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00
26 PCT PZ BY EXERCISE	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
27 PCT PZ BY EXERCISE	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
28 PCT DUES/FEES/CONTINGENT	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
29 OVERTIME MANHOURS USED (21)	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
30 MANHOURS PER FLYING HOUR	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67
31 MOST TRAIL/ISSUE FEES/ ITEMS	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00

	TOTAL	6240	6310	6311	6312	6313	6320	6321	6322	6323	6324	6325	6326
17 MANHOURS AVAILABLE (17)	1270.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
18 MANHOURS AVAILABLE (18)	1270.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
19 PERCENT UTILIZATION	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00
20 MANHOURS USED (19)	1316.01	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4
21 PCT DUES/FEES/CONTINGENT	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
22 PCT SCHED. MAINTENANCE	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
23 NUMBER OF PZ PLANNING	1358.70	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00	474.00
24 NUMBER OF PZ AVAILABLE (20)	1316.01	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4	451.4
25 PCT AVAILABLE (20)	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00	100.00
26 PCT PZ BY EXERCISE	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
27 PCT PZ BY EXERCISE	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
28 PCT DUES/FEES/CONTINGENT	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
29 OVERTIME MANHOURS USED (21)	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
30 MANHOURS PER FLYING HOUR	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67	4.67
31 MOST TRAIL/ISSUE FEES/ ITEMS	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00

J P E R A T I O N S	TOTAL	CONVE	CUMVTF	AAT	AATF	PHASSTF	WASMF	WASHTF	OPLTIF	PELTTF
1. NUMBER OF MESSAGES REQUESTED	2332.00	318.00	134.00	017.00	350.00	86.00	52.00	56.97	28.00	70.00
2. NUMBER ACCOMPLISHED	2229.00	279.00	127.00	055.00	346.00	84.00	42.00	56.92	28.00	70.00
3. PERCENT ACCOMPLISHED	35.56	86.54	79.79	93.96	78.85	100.00	100.00	100.00	100.00	100.00
4. NUMBER OF SORTIES REQUESTED	5894.00	586.00	770.00	190.00	600.00	84.00	42.00	56.92	28.00	70.00
5. NUMBER ACCOMPLISHED	5339.00	571.00	743.00	179.00	545.00	86.00	42.00	56.92	28.00	70.00
6. PERCENT ACCOMPLISHED	90.58	90.31	96.69	93.69	91.11	100.00	100.00	100.00	100.00	100.00
A I R C R A F T										
7. NUMBER OF AIRCRAFT AUTH. ITEMS	72.00	48.00	24.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
8. NUMBER OF AIRCRAFT-DAYS AVAIL.	7056.00	714.00	2352.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
9. PCT SORTIES TIME PELTTF	13.50	13.07	12.56	0.00	0.00	0.00	0.00	0.00	0.00	0.00
10. PCT UNASCHED MAINTENANCE	16.74	16.17	16.69	1.00	0.00	0.00	0.00	0.00	0.00	0.00
11. PCT SCHED MAINTENANCE	21.41	21.40	20.61	0.00	0.00	0.00	0.00	0.00	0.00	0.00
12. PCT REFS	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
13. PCT SERVICE & MEN. WAIT	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
14. PCT OPERATIONALY READY	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
15. AVG. TIME TO GET TO READY TIME	0.72	0.72	0.72	0.00	0.00	0.00	0.00	0.00	0.00	0.00
16. AVG. NO. OF SORTIES/AVL /DAY	0.74	0.75	0.72	0.00	0.00	0.00	0.00	0.00	0.00	0.00
P E R F O R M A N C E										
17. MESSAGE AUTHORIZED (10)	3517.00	133.00	109.76	54.84	47.00	54.80	47.00	47.00	101.92	47.00
18. MESSAGE AVAILABLE (10)	3517.00	133.00	109.76	54.84	47.00	54.80	47.00	47.00	101.92	47.00
19. PERCENT UTILIZATION	3.57	55.49	55.74	52.94	59.65	57.93	0.00	0.00	0.00	0.00
20. MESSAGE USE (10)	1255.00	73.96	61.21	20.05	28.00	31.70	0.00	0.00	0.00	0.00
21. PCT UNASCHED MAINTENANCE	65.00	100.00	99.34	100.00	100.00	100.00	100.00	100.00	100.00	100.00
22. PCT SCHED MAINTENANCE	34.77	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
23. NUMBER OF PTH DEMANDS	1439.00	1304.00	33.00	0.00	1826.00	19.00	0.00	0.00	1470.00	252.00
24. PCT AVAILABLE (10)	73.26	1.76	32.24	0.00	17.76	36.15	0.00	0.00	0.00	0.00
25. PCT AVAILABLE (JUST)	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
26. PCT PROV. BY EFFORT	6.34	0.00	1.27	0.25	1.26	0.00	0.00	0.00	0.00	0.00
27. PCT PROV. BY PREFERENCE	1.12	0.00	1.44	0.99	7.34	2.20	0.00	0.00	0.00	0.00
28. PCT C-130'S NOT SORTED	19.27	48.26	65.00	55.58	73.86	61.27	0.00	0.00	0.00	0.00
29. OVERTIME MESSAGES USED (10)	2.12	0.00	0.34	0.00	0.00	0.00	0.00	0.00	0.00	0.00
30. MESSAGES PER FLYING HOUR	5.00	3.32	0.27	0.13	0.12	0.15	0.00	0.00	0.00	0.00
31. MOST REQUIREMENTS PER HOUR	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00

Figure C-4. Constrained Simulation Performance Summary Report

	P	F	S	C	M	E	L	TOTAL	423X	424X	431C1	431F1	431R1	432T1	432E1	462E1	462E2
27	MANHOUS AUTHORIZED (10)	3517.23	47.4	96.6	67.4	117.6	56.4	47.4	149.35	67.60	47.40	47.40	47.40	47.40	47.40	47.40	47.40
28	MANHOUS AVAILABLE (10)	3517.23	47.4	96.6	67.4	117.6	56.4	47.4	149.35	67.60	47.40	47.40	47.40	47.40	47.40	47.40	47.40
29	PERCENT UTILIZATION	3.57	50.77	44.73	0.10	50.02	50.77	0.22	50.12	50.35	0.57	0.57	0.57	0.57	0.57	0.57	0.57
30	MANHOUS USED (10)	1256.61	21.06	45.13	0.73	59.02	231.67	10.52	79.12	213.84	7.37	7.37	7.37	7.37	7.37	7.37	7.37
31	PCT UNASSED. MAINTENANCE	65.23	06.35	1.00	0.00	24.00	1.00	0.00	1.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
32	PCT SCHEM. MAINTENANCE	34.77	3.65	0.00	11.79	1.00	76.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
33	NUMBR. OF MEN DEMANDED	1439.93	1185.53	559.73	429.02	475.01	587.01	179.73	512.00	463.50	2915.00	404.00	404.00	404.00	404.00	404.00	404.00
34	PCT AVAILABLE (PWRMS)	73.24	46.06	54.74	1.00	26.31	66.01	1.00	7.00	7.00	7.00	7.00	7.00	7.00	7.00	7.00	7.00
35	PCT AVAILABLE (SURST)	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
36	PCT FLOW. BY EXPEDIT	5.16	1.59	1.84	0.00	1.00	14.77	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
37	PCT FLOW. BY PREFERENCE	1.12	3.15	3.71	0.00	0.00	7.84	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
38	PCT DEMANDS NOT SATIS.	19.27	49.30	40.60	0.00	73.69	10.37	0.00	15.11	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
39	OVERTIME MANHOUS USED (10)	2.12	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
40	MANHOUS PER FLYING HOUR	5.46	0.10	0.2	0.04	0.26	1.45	0.05	0.05	0.05	0.05	0.05	0.05	0.05	0.05	0.05	0.05
41	POST TOWLESOME PERS. ITEM	6.12	6.03	7.03	7.02	7.02	7.03	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
TOTAL		3517.23	47.4	96.6	67.4	117.6	56.4	47.4	149.35	67.60	47.40	47.40	47.40	47.40	47.40	47.40	47.40
17	MANHOUS AUTHORIZED (10)	3517.23	47.4	96.6	67.4	117.6	56.4	47.4	149.35	67.60	47.40	47.40	47.40	47.40	47.40	47.40	47.40
18	MANHOUS AVAILABLE (10)	3517.23	47.4	96.6	67.4	117.6	56.4	47.4	149.35	67.60	47.40	47.40	47.40	47.40	47.40	47.40	47.40
19	PERCENT UTILIZATION	3.57	50.77	44.73	0.10	50.02	50.77	0.22	50.12	50.35	0.57	0.57	0.57	0.57	0.57	0.57	0.57
20	MANHOUS USED (10)	1256.61	21.06	45.13	0.73	59.02	231.67	10.52	79.12	213.84	7.37	7.37	7.37	7.37	7.37	7.37	7.37
21	PCT UNASSED. MAINTENANCE	65.23	06.35	1.00	0.00	24.00	1.00	0.00	1.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
22	PCT SCHEM. MAINTENANCE	34.77	3.65	0.00	11.79	1.00	76.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
23	NUMBR. OF MEN DEMANDED	1439.93	1185.53	559.73	429.02	475.01	587.01	179.73	512.00	463.50	2915.00	404.00	404.00	404.00	404.00	404.00	404.00
24	PCT AVAILABLE (PWRMS)	73.24	46.06	54.74	1.00	26.31	66.01	1.00	7.00	7.00	7.00	7.00	7.00	7.00	7.00	7.00	7.00
25	PCT AVAILABLE (SURST)	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
26	PCT FLOW. BY EXPEDIT	5.16	1.59	1.84	0.00	1.00	14.77	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
27	PCT FLOW. BY PREFERENCE	1.12	3.15	3.71	0.00	0.00	7.84	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
28	PCT DEMANDS NOT SATIS.	19.27	49.30	40.60	0.00	73.69	10.37	0.00	15.11	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
29	OVERTIME MANHOUS USED (10)	2.12	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
30	MANHOUS PER FLYING HOUR	5.46	0.10	0.2	0.04	0.26	1.45	0.05	0.05	0.05	0.05	0.05	0.05	0.05	0.05	0.05	0.05
31	POST TOWLESOME PERS. ITEM	6.12	6.03	7.03	7.02	7.02	7.03	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
TOTAL		3517.23	47.4	96.6	67.4	117.6	56.4	47.4	149.35	67.60	47.40	47.40	47.40	47.40	47.40	47.40	47.40

Figure C-4. Constrained Simulation Performance Summary Report (continued)

TASKS	THIS MATRIX COVERS 60 DAYS OF SIMULATION DATA												
1	1	1	1	1	1	1	1	1	1	1	1	1	1
2	1	1	1	1	1	1	1	1	1	1	1	1	1
3	1	1	1	1	1	1	1	1	1	1	1	1	1
4	1	1	1	1	1	1	1	1	1	1	1	1	1
5	1	1	1	1	1	1	1	1	1	1	1	1	1
6	1	1	1	1	1	1	1	1	1	1	1	1	1
7	1	1	1	1	1	1	1	1	1	1	1	1	1
8	1	1	1	1	1	1	1	1	1	1	1	1	1
9	1	1	1	1	1	1	1	1	1	1	1	1	1
10	1	1	1	1	1	1	1	1	1	1	1	1	1
11	1	1	1	1	1	1	1	1	1	1	1	1	1
12	1	1	1	1	1	1	1	1	1	1	1	1	1
13	1	1	1	1	1	1	1	1	1	1	1	1	1
14	1	1	1	1	1	1	1	1	1	1	1	1	1
15	1	1	1	1	1	1	1	1	1	1	1	1	1
16	1	1	1	1	1	1	1	1	1	1	1	1	1
17	1	1	1	1	1	1	1	1	1	1	1	1	1
18	1	1	1	1	1	1	1	1	1	1	1	1	1
19	1	1	1	1	1	1	1	1	1	1	1	1	1
20	1	1	1	1	1	1	1	1	1	1	1	1	1
21	1	1	1	1	1	1	1	1	1	1	1	1	1
22	1	1	1	1	1	1	1	1	1	1	1	1	1
23	1	1	1	1	1	1	1	1	1	1	1	1	1

AVE DLMR 2.3 2.7 2.1 2.1 1.7 2.1 1.9 1.7 1.5 1.4 1.6 1.8 2.2 1.9 2.2 2.1 2.3 2.4 2.3 2.1 2.4
 ST DEV/HR 2.3 2.7 2.1 2.1 1.7 2.1 1.9 1.7 1.5 1.4 1.6 1.8 2.2 1.9 2.2 2.1 2.3 2.4 2.3 2.1 2.4
 TOTAL TASKS B/D FOR LACK OF MPMR = 116 AVE TASKS P/D / DAY = 1.95 STC DEV / DAY = 2.97 TOTAL MFS CF DELAY = 556.94
 TOTAL MEN P/D = 148 AVE MEN P/D / TSM = 1.3 STC DEV = .45 AVE MEN ST REQST = 1.6 AVE DELAY PER TASK = 4.82
 TOTAL REQUIRED MANNING = 1.00
 SHIFT NUMBER 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23
 START MOUT-STOP HOUR 07 16 16 24
 DISPECT MANNING/TOTAL MANN INDIPECT = 1.74 7 2.89 1.16 7 1.97 1.77 7 2.84
 MAXIMUM CREW SIZE PER JOB 2

Figure C-6. AFSC 531X3 off-Equipment Constrained Manpower Backorder Matrix

APPENDIX D

FAILURE CLOCKS

APPENDIX D

FAILURE CLOCKS

This appendix lists the LCOM failure clocks, the respective mean sorties between maintenance action (MSBMA), the decrement tasks, and the decrement values. The MSBMA's are in aircraft sorties unless otherwise specified. DCRMT1, DCRMT2, and DCRMT5 are located in the main flight line networks. DCRMT1 advances the failure clocks prior to a mission launch while DCRMT2 and DCRMT5 advance the clocks after launch. All other decrement tasks are located in the phase and/or corrective maintenance networks.

The cumulative decrement values corresponding to certain failure clocks listed in Table D-I do not equal one. This apparent inconsistency is corrected in respective corrective maintenance network. Tetmeyer (Ref 33) describes in detail the procedures used in designing the corrective maintenance networks and decrementing the corresponding failure clocks.

Table D-I
Failure Clocks

LCOM CLOCK	MSBMA	DECREMENT	
		DCRMT1	DCRMT2
F11A00	18	.02	.98
F11D00	6	.02	.98
F11G00	8	.02	.98
F11K00	20	.02	.98
F11P00	22	.02	.98
F12A00	102	.02	.98
F12B00	125	.02	.98
F12C00	25	.02	.98
F13A00	11	.09	.91
F13B00	26	.09	.91
F13C00	164	.09	.91
F13D00	65	.09	.91
F13F00	178	.09	.91
F13H00	237	.09	.91
F13J00	19	.09	.91
F14AA0	93	.05	.95
F14AB0	711	.05	.95
F14C00	152	.05	.95
F14D00	21	.05	.95
F14E00	67	.05	.95
F14G00	533	.05	.95
F14H00	426	.05	.95
F23000	5	.06	.94
F23100	42	.06	.94
F24A00	44	.14	.86
F24B00	26	.14	.86
F24D00	355	.14	.86
F27000	5	.06	.94
F41A00	20	.05	.95
F42A00	67	.16	.84
F44A00	20	.09	.91
F44B00	267	.09	.91
F44E00	305	.09	.91
F45A00	112	.07	.93
F45B00	133	.07	.93
F45C00	23	.07	.93
F46A00	32	.06	.94
F46B00	355	.06	.94
F46D00	305	.06	.94
F46E00	32	.06	.94
F47A00	63	.10	.90
F49A00	85	.06	.94
F51A00	50		.95
F51E00	51		.95
F51L00	15	.05	
F51M00	426		.95
F51N00	27		.95

Table D-I. Failure Clocks (continued)

LCOM CLOCK	MSBMA	DECREMENT	
		DCRMT1	DCRMT2
F52A00	35	.19	.81
F55A00	237		.99
F55B00	65		.99
F55C00	76		.99
F55L00	420	.01	
F57A00	21		.87
F57L00	70	.13	
F63A00	27		.91
F63B00	25		.91
F63L00	12	.09	
F65A00	142		.98
F65B00	58		.98
F65L00	69	.02	
F71A00	13		.93
F71B00	305		.93
F71C00	533		.93
F71D00	55		.93
F71F00	52		.93
F71L00	8	.07	
F74E00	164		.93
F74F00	4		.93
F74J00	36		.93
F74K00	18		.93
F74L00	6	.07	
F75B00	51		1.00
F75C00	185		1.00
F75D00	152		1.00
F75F00	107		1.00
F75M00	213		1.00
F76A00	18	.11	.89
F76G00	250	.11	.89
F76H00	12		1.00
FCTANK	1333		1.00
FS11PA	590		1.00
FS13HC	1500		1.00
FS14GC	590		1.00
FS2300	33		1.00
FS23HA	89		1.00
FS4700	60	<u>DCRMT3</u> 1.00	
FS7500	30	1.00	

Table D-I. Failure Clocks (continued)

LCOM CLOCK	MSBMA	DECREMENT		
F75H00 FSCGUN FSGUN0	20000 rounds 15000 rounds 25000 rounds	<u>DCRMT5</u> 41.40 rounds 41.40 rounds 41.40 rounds		
FD6000	9	<u>DCRMG2</u> .30	<u>DCRMG3</u> .30	<u>DCRMG7</u> .40
FTTU00	5	<u>DCRMH7</u> .70		
FTSC	65 demands	<u>DCRMTA</u> 1.00 demand		
FTSD	65 demands	<u>DCRMTB</u> 1.00 demand		
FTSM	65 demands	<u>DCRMTC</u> 1.00 demand		
HF	1	<u>DCRMF</u> 1.00		
HTF	1	<u>DCRMTF</u> 1.00		

APPENDIX E

SPARE PART AND AVIONIC TEST STATION CONSTRAINTS

Table E-I. Spare Part Constraints

Work Unit Code	Constrained Quantity	Work Unit Code	Constrained Quantity
13CA0	11	52AB0	7
14AA0	8	52AC0	14
14AB0	1	52AL0	8
23000	37	52AM0	10
24AA0	14	55AC0	3
24B00	31	55AD0	4
51AD0	16	55BC0	5
51AE0	23	55BE0	3
51AF0	11	55CA0	6
51AG0	14	55CB0	33
51AH0	14	57AA0	6
51AK0	18	63AG0	11
51AL0	14	63BC0	14
51AM0	9	63BD0	8
51EA0	4	63BF0	8
51ED0	9	63BH0	7
51NA0	6	65AA0	8
51NB0	10	65BH0	8
52AA0	7	71AE0	14

AD-A156 548

ESTIMATION OF F-15 PEACETIME MAINTENANCE MANPOWER
REQUIREMENTS USING THE (U) AIR FORCE INST OF TECH
WRIGHT-PATTERSON AFB OH SCHOOL OF ENGI.

3/3

UNCLASSIFIED

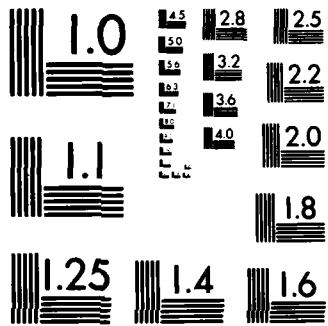
G DEGOVANNI ET AL. DEC 76 AFIT/GOR/SM/76D-5 F/G 5/9

NL

END

FILED

DTIC



MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

Table E-I. Spare Part Constraints (Continued)

Work Unit Code	Constrained Quantity	Work Unit Code	Constrained Quantity
71AKO	20	74FKO	16
71BDO	6	74FQO	12
71CAO	7	74FSO	13
71FAO	9	74FUO	11
71FBO	10	74JAO	12
71FCO	11	74JCO	11
71FEO	9	74KAO	7
74EBO	12	74KCO	8
74FAO	14	74KEO	7
74FCO	16	75BBO	5
74FFO	14	75HEO	7
74FHO	16	75MAO	7
74FJO	14	75MCO	5

Table E-II
Avionic Test Station Constraints

Avionic Test Station	Job Description	Constrained Quality
TSC	Analog/Digital Computer Test Station	2
TSD	Analog/Visual Computer Test Station	2
TSM	Microwave Frequency Test Station	2

VITA

George DeGovanni was born on 4 January 1947 in Philadelphia, Pennsylvania. He graduated from the United States Air Force Academy in June 1968 with a Bachelor of Science in Mathematics. He completed Undergraduate Pilot Training at Moody AFB, Georgia in August 1969. He then flew combat missions in Vietnam as an OV-10 Forward Air Controller attached to the 20th Tactical Air Support Squadron until April 1971. Upon return to the United States, DeGovanni was assigned to the 14th Pilot Training Wing, Columbus Air Force Base, Mississippi, as a T-38 Instructor Pilot/Flight Examiner. In May 1975, he entered the Air Force Institute of Technology.

Permanent Address: 367 Leroy Street
Philadelphia, Pennsylvania 19128

VITA

Donald Michael Douglas was born on 23 January, 1943 in Mt. Carmel, Illinois. After graduating from high school there in 1961, he attended the United States Air Force Academy from which he received the degree of Bachelor of Basic Science and a commission in the United States Air Force in 1965. He completed pilot training in September, 1966 and served as a KC-135 pilot with the 301st Air Refueling Wing, Lockbourne Air Force Base, Ohio. After completing rotary wing conversion training in May, 1969, he served with the 809th Combat Support Group, F. E. Warren Air Force Base, Wyoming as a UH-1F pilot. During 1970 and 1971 he flew as a UH-1P and UH-1N gunship flight examiner with the 20th Special Operations Squadron, RVN. He then served as an instructor pilot and Wing Chief of UH-1N Standardization in the 1550th Aircrew Training and Test Wing, Hill Air Force Base, Utah until entering the School of Engineering, Air Force Institute of Technology, in May 1975.

Permanent address: 315 Walnut Street
Mt. Carmel, Illinois 62863

END

FILMED

8-85

DTIC