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PARTIAL SUBSTITUTION AND OTHER MODIFICATIONS TO THE PARCOM MODEL (SHORT TITLE: PARCOM PARTIAL SUBSTITUTION)

NOVEMBER 1984



PREPARED BY FORCE SYSTEMS DIRECTORATE US ARMY CONCEPTS ANALYSIS AGENCY 8120 WOODMONT AVENUE BETHESDA, MARYLAND 20814-2797

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a full-sub set, within which all installed serviceable parts on NMCS aircraft are substitutable for unavailable spares, and a no-sub set, within which no installed parts are substitutable for spares. The extended PARCOM was applied in several illustrative example cases, showing plausible results as substitution policy was varied.

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NOVEMBER 1984

PREPARED BY FORCE SYSTEMS DIRECTORATE US ARMY CONCEPTS ANALYSIS AGENCY 8120 WOODMONT AVENUE BETHESDA, MARYLAND 20814-2797



PARTIAL SUBSTITUTION AND OTHER MODIFICATIONS TO THE PARCOM MODEL (PARCOM PARTIAL SUBSTITUTION)

STUDY SUMMARY CAA-TP-84-11

THE REASON FOR PERFORMING THE STUDY was that the two models recommended by the Aircraft Spares Study, Overview and PARCOM, could treat a fullsubstitution or a no-substitution part replacement policy but lacked the ability to represent a more realistic partial-substitution replacement policy. Of the two models, PARCOM was judged to be the better candidate for incorporation of a partial-substitution capability.

THE PRINCIPAL FINDINGS of the work reported herein are as follows:

(1) The basic PARCOM (Parts Requirements and Cost Model), developed for the Aircraft Spares Study, was extended to include the effects of partial-substitution replacement policies and deployment of initial stocks over time.

(2) The resulting extended model relates spare requirements to a flying hour/aircraft availability objective, parts replacement policy, and stockage deployment schedule--all subject to optional cost constraints. Example applications illustrated the plausibility of the model logic.

(3) The extended PARCOM significantly expands the range of application and results of the basic PARCOM methodology. As such, its implementation, in place of basic PARCOM, is warranted.

THE MAIN ASSUMPTION was that partial substitution can be usefully defined in terms of a partition of part types into a full-substitution part set and a no-substitution set.

THE PRINCIPAL LIMITATION was that definitions of partial substitution other than the assumed definition might not be addressable by the extended PARCOM.

THE SCOPE OF THE STUDY addressed the relationship of spare requirements and fleet capability for a notional Army aviation program to a flying hour/availability objective, part replacement (substitution) policy, and stockage deployment schedule--all subject to optional cost constraints. The study applied the subject model to an example, using four part types over 5 days, and to an all-up case, treating an AH-1S scenario involving 334 part types over 120 days.

THE STUDY OBJECTIVES were:

(1) To evaluate the potential for extending the capability of the basic PARCOM, developed in the Aircraft Spares Study, to include partial substitution and other desirable features lacking in the basic PARCOM.

(2) To make the above extensions and to report on and illustrate the application of the extended PARCOM and methodology.

THE BASIC APPROACH was:

(1) To assess the limitations of the basic PARCOM.

(2) To select features and capabilities, to include partial substitution, for incorporation into an extended PARCOM.

(3) To develop an extended PARCOM incorporating the selected capabilities.

(4) To report on the nature of the extended PARCOM methodology and model through exposition and illustrative example applications.

THE STUDY SPONSOR was the Deputy Chief of Staff for Logistics, Headquarters, Department of the Army.

THE STUDY EFFORT was conducted by Mr. Walter J. Bauman, Force Systems Directorate, US Army Concepts Analysis Agency.

COMMENTS AND QUESTIONS may be directed to the Director, US Army Concepts Analysis Agency, ATTN: CSCA-FS, 8120 Woodmont Avenue, Bethesda, Maryland 20814-2797.

Tear-out copies of this synopsis are at back cover.

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STUDY SUMMARY (tear-out copies)

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PARTIAL SUBSTITUTION AND OTHER MODIFICATIONS TO THE PARCOM MODEL (Short title: PARCOM Partial Substitution)

CHAPTER 1

MODEL DEVELOPMENT

1-1. BACKGROUND

a. Model Origin. The US Army Concepts Analysis Agency (CAA) developed the Parts Requirements and Cost Model (PARCOM) to generate cost effective mixes of aircraft spare parts and to assess aircraft fleet performance under specified wartime scenario conditions. Development occurred during the course of the Aircraft Spare Stockage Methodology (Aircraft Spares) Study1 conducted by CAA. That study, and PARCOM development, were in response to interest shown by the Deputy Chief of Staff for Logistics (DCSLOG) in devel-oping a methodology (or methodologies) relating aircraft spare parts stockage levels to combat readiness and flying hour capability. The calculation of spare parts requirements, and of the effects of budgeting changes, had been a slow and cumbersome peacetime oriented exercise. The principal criterion for spares stockage had been the achievement of acceptable stockout, or fill rate, levels. To more realistically predict wartime spare parts requirements, and to better justify budget requests for spare parts procurement, the Army needed a more responsive methodology based on wartime flying hour expectations and system readiness/availability requirements. PARCOM was developed to meet that need.

b. **Current Study Purpose.** Results reported in Aircraft Spares were sufficiently encouraging to warrant a follow-on study, designated the Overview/PARCOM Turnkey Project (OPTP). Included in the objectives of OPTP were the following actions pertaining to PARCOM:

(1) Document the PARCOM, as developed in the Aircraft Spares Study, and deliver it to the US Army Aviation Systems Command (USAVSCOM).

(2) Evaluate and report on the potential for extending the capability of PARCOM to include partial-substitution parts replacement policies and any other features deemed desirable, but lacking in the model (PARCOM) developed for Aircraft Spares.

This technical paper is a report on the model extension. The extended model reported herein is denoted as extended PARCOM; the original model, as developed in the Aircraft Spares Study, is denoted herein as basic PARCOM.

1-2. BASIC PARCOM PROBLEM SPECIFICATION. The basic PARCOM was designed to generate cost-effective mixes of add-on spare parts needed to permit an aircraft fleet of specified type to achieve specified flying program and availability goals under various cost constraints, part replacement policies, and aircraft availability objectives. These are described below in summary fashion. Additional detail may be found in the PARCOM User's Guide.²

a. Cost Constraints. The two cost constraint modes are:

(1) Unconstrained Funds - where unlimited funds for procurement of additional required parts are assumed available.

(2) Constrained Funds - where a cost (funding) limit for add-on spares is set. If unable to meet the flying hour and, possibly, availability objectives with the limited funds, the model generates a "best" solution mix with the funds available, i.e., it seeks to maximize program flying hours achievable within the funding constraint.

b. Basic Part Replacement Policies. The two basic part replacement policies are:

(1) Full Substitution - where a failed part on an aircraft may be replaced by either a spare (if available) or by a serviceable part installed on a not-mission-capable (NMC) aircraft (if a spare is not available).

(2) No Substitution - where a failed part on an aircraft may only be replaced by a spare part.

c. Flying Hour Objective. A flying hour objective is a requirement for the aircraft fleet to achieve a specified number of flying hours on each day of the scenario. An input flying hour program designates the daily goal. A basic PARCOM objective is to generate a parts mix which will achieve the specified flying program at least cost.

d. Aircraft Availability Objective. An aircraft availability objective is a requirement for a specific minimum aircraft availability on each day (different days may have different minimum required availabilities). In this context, aircraft availability = 1 - NMCS, where NMCS = the fraction of surviving aircraft in not-mission-capable-supply status. An aircraft is in an NMCS status if it is nonoperational because spare parts are needed but are not available to restore it to serviceability. Specification of availability objectives is in addition to the flying hour objective. Specification of a zero availability objective is equivalent to no availability objective at all.

1-3. SUMMARY OF REQUIREMENTS OUTPUT FOR BASIC PARCOM. The following are the types of print output produced by basic PARCOM for requirements problems. Details may be found in the PARCOM User's Guide.

a. Unconstrained Cost Cases

(1) Total Requirement. Total least-cost parts mix and costs required to achieve the case objectives (flying program and availability) given a zero initial inventory.

(2) Residual Requirement. The least-cost add-on parts mix (to an input initial inventory) and costs required to achieve the case objectives.

(3) Cumulative Cost by Day. For each day N (N=1, 2, ..., through end of war), the total and the add-on costs of the full parts requirements to meet the case objectives through day N only, i.e., it is the cost of the requirement for a truncated scenario of N days. Parts mix is not shown.

(4) Cumulative Requirement by Day. For selected items, for each day N, the cumulative total parts requirement needed (in the full parts scenario) to meet the case objectives through N days. A zero initial inventory is assumed in this output.

(5) Daily Aircraft Available. For each day of the full scenario, the fraction of surviving aircraft which are <u>not</u> NMCS, assuming that the starting spare inventory is set equal to the sum of the computed parts requirement and the initial inventory.

(6) Daily Flying Hours per Aircraft per Day. For each day of the scenario, the average achieved flying hours per available aircraft per day assuming the computed solution parts mix is stocked.

b. Constrained Costs

(1) Total Requirement. Total best requirements mix, with zero initial inventory, and with a no-substitution policy, that can be bought with a funding limit equal to the sum of the values of the current spares inventory and the input cost limit. The objective of a best mix is to maximize flying hour productivity with the constrained funds.

(2) Residual Requirement. Add-on (to input initial inventory) requirements mix, with a no-substitution policy, that can be bought with a funding limit equal to the input cost limit.

(3) Daily Aircraft Available. For each day of the full scenario, the fraction of surviving aircraft which are <u>not</u> NMCS, assuming that the starting spare inventory is set equal to the sum of the computed parts requirement and the initial inventory.

(4) Daily Flying Hour Fraction. For each day of the full scenario, the fraction of the fleet flying program which can be achieved assuming that the starting spare inventory is set equal to the sum of the computed parts requirement and the initial inventory.

(5) Daily Flying Hours per Aircraft per Day. For each day of the scenario, the average achieved flying hours per aircraft per day, assuming the computed solution parts mix is stocked.

1-4. LIMITATIONS OF BASIC PARCOM. The following limitations of the basic PARCOM were noted as a suitable base for future model extension and/or redesign.

a. No Partial-substitution Requirements. A partial-substitution parts replacement policy can be conceptualized as one in which some (but not necessarily all) part types installed in NMCS aircraft are substitutable

for spares, i.e., such a part type installed in an NMCS aircraft can, if serviceable, be applied to replace a failed part (of the same type) in another NMCS aircraft if a spare is unavailable. The basic PARCOM does not consider partial substitution. The basic PARCOM requirement algorithms process only a full-substitution replacement policy (all parts substitutable) or a no-substitution replacement policy (no parts substitutable) depending on the case treated.

(1) Unconstrained Cost. The basic PARCOM calculates unconstrained cost requirements with both full substitution and no substitution. However, using a common scenario, PARCOM-generated solution requirement costs under a no-substitution policy are much larger (for nontrivial cases) than solution costs under full substitution. It may be useful, therefore, to examine the effects of partial-substitution policies with corresponding intermediate solution costs.

(2) Constrained Cost. While the standard unconstrained cost requirements solution of the basic PARCOM can treat both full substitution and no substitution, the constrained cost algorithm of that model treats only a no-substitution replacement policy. Extension to processing of partial substitution would enhance model capability.

b. No Partial-substitution Fleet Capability Assessment. The basic PARCOM assesses fleet flying capability (resulting aircraft availability and fraction flying program achieved) based on a solution inventory being stocked or on a current (input-specified) parts inventory. Capability assessments based on unconstrained cost solutions treat both full substitution and no substitution, but assessments based on constrained cost solutions or on a current inventory treat only a no-substitution policy. Application of full substitution and no substitution produce upper and lower bounds, respectively, on assessments of fleet flying hour capability with a fixed spare inventory. Modeling of partial substitution will enable a cause-and-effect analysis of flying hour capabilities between those bounds.

c. No Parts Distributed Over Time. The basic PARCOM, in both assessment and requirement calculations, assumes that all initial spare assets are "front-loaded," i.e., that all initial spares are available at retail on Day 1 of the scenario. Since a spare has no effect unless it is needed, this is equivalent to assuming that all initial assets will reach retail before they are required (as replacements). An efficient stockage and transportation system will achieve this. However, some scenarios will not be optimally matched to the time-phased parts deployments reflected in the authorized stockage list (ASL)/prescribed load list (PLL) of deploying units and the depot-retail pipeline lag for stocks initially at depots. The basic PARCOM, therefore, may yield overly optimistic results. Greater credibility and conformance to real-life constraints can be achieved by enabling PARCOM to process time-phased parts deployments.

1-5. PARCOM EXTENSION

a. Need. The objective of the Aircraft Spares Study was only to develop and demonstrate a feasible methodology. The follow-on study effort, OPTP, was to document the basic PARCOM and to deliver it (as well as Overview) to the US Army Aviation Systems Command (USAVSCOM). The selected models included the basic PARCOM; however, OPTP also proposed to study means of extending the basic PARCOM to include partial substitution and other capabilities found feasible and useful. The final OPTP report was to include an evaluation of the feasibility of implementing these model extensions. This technical paper presents that evaluation.

b. Aspects Selected for Extension. The basic PARCOM limitations noted in paragraph 1-4 were chosen as the basis for extending PARCOM capability, i.e., the extended PARCOM was designed to have the capability to analyze:

(1) Effects of using partial-substitution part replacement policies in requirements calculations.

(2) Effects of using partial-substitution part replacement policies in fleet capability assessment.

(3) Effects of using input-specified parts deployments over time.

(4) Effects of cost constraints on requirements solutions using partial substitution.

In the above context, partial substitution includes full substitution (full sub) and no substitution (no sub) as special cases.

c. Methodology. The approach to PARCOM extension included:

(1) Selection of the capabilities to be added. These are noted above.

(2) Construction of a concept for partial substitution amenable to processing in an extended PARCOM.

(3) Revision or replacement of program code in basic PARCOM to enable demonstration of concept feasibility for the extensions.

(4) Checking, via selected manual examples, or all-up tests, of concept feasibility for the extensions.

(5) Provision of an undocumented copy of the FORTRAN program source code for the extended PARCOM.

1-6. FUTURE OUTPUT. The products of the OPTP do not include delivery and documentation of a complete extended PARCOM (only the basic PARCOM is delivered and documented). However, a follow-on effort of limited scope will provide:

a. Publication of revisions to the (basic) PARCOM User's Guide and Functional Description.

b. Documentation of the program source code for the extended PARCOM.

CHAPTER 2

REQUIREMENTS DETERMINATION WITH PARTIAL SUBSTITUTION

2-1. CONCEPT FORMULATION

a. Definition. In the extended PARCOM, a partial-substitution parts replacement policy is defined by partitioning all part types into a full-sub set and a no-sub set. A part type is in only one set and remains in that set throughout the scenario. These sets are defined as follows:

(1) All parts in the full-sub set operate with a full-substitution replacement policy relative to aircraft which are NMCS due to lack of a part from that set. That is, a failed full-sub part on an aircraft may be replaced either by a spare (if available) or by a serviceable part installed on an NMCS aircraft which is awaiting a full-sub part, if a spare is not available. However, no failed full-sub part can be replaced by any part installed on an NMCS aircraft awaiting a no sub-part.

(2) Parts in the no-sub set operate with a no-substitution replacement policy. That is, a failed no-sub part on an aircraft may only be replaced by a spare part. An NMCS aircraft lacking a no-sub part may neither receive a serviceable part from another NMCS aircraft, nor may it provide a service-able part to (fill a "hole" in) any other NMCS aircraft.

b. Implications. The full-substitution and no-substitution policies of the basic PARCOM are special cases of partial substitution in which all parts are either in the full-sub set or in the no-sub set. The analytic usefulness of the above definition arises from the consequence that any NMCS aircraft will either be awaiting exactly one no-sub part or at least one full-sub part but will never be awaiting a mixture of full-sub and no-sub parts.

c. Selection of Full-sub Parts. Before requirements processing begins in the extended PARCOM, a full-sub and a no-sub part set, applicable over all scenario days, must be defined. One option allows the user to specify those part types which comprise the full-sub set. By default, all nonspecified parts are presumed to be in the no-sub set. However, the model has another option, allowing the user to specify three screening limits--L1, L2, and L3. With these limits the model selects a part type for the fullsub set if at least one of the following apply:

- The (input) depot repair cycle time for the part exceeds L1 days and the not repairable this station (NRTS) fraction exceeds zero.
- The (input) NRTS fraction for the part exceeds L2.
- The (input) retail repair time for the part exceeds L3.

The model, under this option, examines all part types and assigns those that satisfy the screening limits to the full-sub set. All other part types are assigned to the no-sub set. The above screening criteria were chosen because it appeared plausible that full substitution would be most likely practiced on parts which took a long time to cycle back through the repair pipelines; however, other criteria could also be selected.

2-2. EXAMPLE TEST. The application of the partial-substitution concept is demonstrated below via illustrative examples. For review and comparison purposes, the effects of standard (basic PARCOM) full-substitution and no-substitution policies are also shown, followed by a summary of partial-substitution logic and effect calculations.

a. Problem Framework

(1) A data base containing four part types was applied in a 5-day scenario. Table 2-1 shows input parts data for the example. QPA denotes the quantity per application, i.e., the number of installed parts per operational aircraft. OST denotes the one-way, order and ship time between depot and retail. Overall repair cycle equals the sum of the depot repair time and 2xOST for depot repairable items and equals the retail repair time for retail repairable items. Essentially, it is the (pipeline) time between removal of a failed part and its return to the retail pool of serviceable spares. Table 2-2 shows scenario input data for the example problem. The two columns on the right define the flying hour and availability objectives for the problem. The cumulative aircraft deployments and losses are also input. Cumulative aircraft surviving is calculated from them.

	PART	L PART	PART	> PART	4
• PARTS CHARACTERISTICS					
- FAILURE RATES (PER FL	Y HR) .08	.02	.06	•02	
- QPA	1	1	1	1	
- UNIT COST (\$)	400	50	40	30	
- INIT (INITIAL STOCK)	250	10	260	30	
• PARTS REPAIR CYCLE DATA					
- OST (DAYS)	1	0	1	0	
- RETAIL REPAIR TIME (D	AYS) 0	3	0	2	
- DEPOT REPAIR TIME (DA	YS) 1	0	2	0	
- RETAIL CONDEMN %	0	0	0	0	
- DEPOT CONDEMN Z	0	0	0	0	
- NRTS Z	100	0	100	0	
(OVERALL REPAIR CYC	(3	DAYS) (3	DAYS) (4	DAYS) (2	DAYS)

Table	2-1.	Example	Problem	-	Parts	Data
IGDIC		Example.	1 I OD I GII		1 41 63	Data

DAY	ACFT DEPL	CUM ACFT LOST	CUM ACFT <u>SURY</u>	FLYING HR PGM (FHP)	MIN ACFT AVAIL
1	150	0	150	500	.10
2	200	0	200	1,000	.09
3	200	0	200	1,000	.09
4	200	0	200	1,500	.09
5	200	0	200	1,500	.09

Table 2-2. Example Problem - Scenario Data

(2) Given the problem input data, Table 2-3 shows necessary preprocessing used in all algorithm calculations. The allowable NMCS aircraft for a day is the maximum number of surviving aircraft which can be NMCS on that day while still allowing fleet accomplishment of the case objective (flying hour and availability) for that day. FHP denotes the specified flying hour program for each day.

Table 2-3. Calculation of Allowable NMCS Aircraft

DAT	MIN ACFT RUR BY FLYING HR <u>PROGRAM</u> FHP/MFHAD*	MIN ACFT ROR BY AVAILABILITY <u>CONSTRAINT</u> SURV AC X MIN AVAIL	UVERALL MIN ACFT <u>REQUIRED</u> MAX (#1, #2)	ALLOWABLE <u>NMCS ACFT</u> SURV AC -#>
1	500/10=50	150x.10=15	50	150-50=100
2	100	18	100	100
3	100	18	100	100
4	150	18	150	50
5	150	18	150	50

b. Unconstrained Cost Requirements Under Full Substitution. Tables 2-4 through 2-7 illustrate the basic PARCOM logic for example data under a full-substitution policy. Formulas used in calculations are shown in the table headings. The allowable stockout for a part on a day is just the maximum number of backorders (unfilled demands) for the part which will still allow accomplishment of the case objectives on that day. The day requirement is the minimum add-on stock required to achieve the objectives on a given day. The largest of all the day requirements for a part (circled in the table) is the overall (minimum) requirement for the part. All stock is assumed front-loaded, i.e., available at retail when needed.

DAY	#1 CUMULATIVE <u>FAILURES</u> (FAIL RT X FHP X QPA)	#2 CUM RET <u>REPAIRS</u>	#3 CUMULATIVE <u>NET DEMAND</u> MAX [#1-(#2+INIT)]	#4 ALLUWABLE <u>STUCKOUTS</u> NMCS AC X QPA	DAY <u>RQMTS</u> MAX (#3-#4)
1	.08×500=40	0	UR (U)	$100 \times 1 = 100$	0R(0) 0-100 UR[0]
2	40+.08x1000=120	0	0	100	0
3	200	0	0	100	Û
4	320	40	320-290=30	50	0
5	440	120	70	50	2

Table 2-4. Unconstrained Cost Residual Requirement with Full Substitution - Part 1 (initial inventory = 250)

PART 1 OVERALL ROMT = LARGEST DAY ROMT = 20

Table 2-5. Unconstrained Cost Residual Requirement with Full Substitution - Part 2 (initial inventory = 10)

UAY	#1 CUMULATIVE FAILURES (FAIL RT X FHP X QPA)	#2 CUM RET <u>KEPAIRS</u>	#3 CUMULATIVE <u>NET DEMAND</u> MAX [#1-(#2+INIT)] UR (0)	#4 Allowable <u>stockouts</u> NMCS AC X QPA	DAY <u>RQMT</u> MAX(#>-#4) OR (O)
1	•02x500=10	0	10-0-10 UR 0	100×1=100	0-100 UR U
2	10+.02x1000=30	0	20	100	0
3	50	0	40	100	0
4	80	10	80-10-10=60	50	60-50=10
5	110	30	70	50	70-50-20

PART 2 UVERALL ROMT = LARGEST DAY ROMT = 20

DAY	#1 CUMULATIVE FAILURES (FAIL RT X FHP X QPA)	#2 CUM RET <u>REPAIRS</u>	#.5 CUMULATIVE <u>NET DEMAND</u> MAX [#1-(#2+INIT)] UR (U)	#4 ALLUWABLE <u>STUCKUUTS</u> NMCS AC X QPA	UAY <u>RQMT</u> MAX(#3-#4) UR (U)
1	30	0	0	100	0
2	90	0	0	100	0
3	150	0	0	100	0
4	240	0	Û	50	0
5	330	30	40	50	0

Table 2-6. Unconstrained Cost Residual Requirement with Full Substitution - Part 3 (initial inventory = 260)

PART 3 UVERALL ROMT = 0

Table 2-7. Unconstrained Cost Residual Requirement with Full Substitution - Part 4 (initial inventory = 30)

VAV	#1 CUMULATIVE FAILURES (FAIL RT X FHP X QPA)	#2 CUM RET <u>REPAIRS</u>	#3 CUMULATIVE <u>NET DEMAND</u> MAX [#1-(#2+INIT)] UR (0)	#4 Alluwable <u>STOCKOUTS</u> NMCS AC X ⊌PA	DAY <u>RQMT</u> MAX(#3-#4) OR (U)
1	10	0	0	100	0
2	30	0	0	100	0
3	50	10	10	100	0
4	80	30	20	50	0
5	110	50	30	50	0
		PART	4 OVERALL ROMT = 0		

c. Unconstrained Cost Requirements Under No Substitution. Tables 2-8 through 2-11 illustrate the basic PARCOM logic for the example under a nosubstitution policy. In this case, requirements must be calculated in order of decreasing part unit cost (i.e., most expensive parts first). For a no-substitution policy, the total allowed stockout consists of the summed stockouts over all parts treated. However, since requirements are calculated (purchased) sequentially, each successive calculation uses an "unallocated allowable stockout" equal to the original (Table 2-3) allowable stockout reduced by the sum total of allocated stockouts reflected in purchases of parts already processed. As before, the overall part requirement (circled) is calculated as the largest of the day requirements.

d. Summary of Full-sub and No-sub Results. Table 2-12 summarizes the results thus far. The full-sub requirement cost is the cheapest over all part replacement policies while the no-sub cost is the most expensive. All partial-sub requirements costs must be between these values. This will be illustrated subsequently.

	CALCULATE FOR	MOST EXPENSIVE PART (P	ART 1)
	#1	#2	
	CUM NET		
	DEMAND	UNALLOCATED	
	PART 1	ALLOWABLE	
DAY	(INIT=250)	STUCKUUT	DAY ROMT
	(FROM 'FULL SUB')	(=ALLOWED NMCS AC)	MAX(#1 - #2) UR (0)
1	0	100	0 - 100 OR (
2	0	100	Û
3	0	100	0
4	30	50	0
	70	50	0

Table 2-8. Unconstrained Cost Residual Requirement with No Substitution - Part 1 (initial inventory = 250)

PART 1 ROMT = LARGEST DAY ROMT = 20

		CALCULATE ROMT FOR ASSUME PREVIOUS (P. INIT = OLD INIT (2	NEXT MUST EXPENS ART 1) RQMT "BOUG 50) + RQMT (20) =	IVE PART (PART 2) HT" SU THAT NEW PART 1 270
	#5	#4	#5	
	CUM NET	CUM NET		
	DEMAND	DEMAND	UNALLOCATED	
	PART 1	PART 2	ALLOWABLE	
DA	(<u>(INII=2/0)</u>	<u>(INII=10)</u>	STOCKOUT	DAY ROMI
	#1-20	(FROM 'FULL SUB')	#2-#3	MAX(#4 - #5)
	UR (U)		UK (U)	UR (U)
1	0	0	100-0=100	0
2	0	20	100-0=100	0
3	0	40	100-0=100	0
4	30-20=10	60	50-10=40	60-40=20
5	70-20=50	70	50-50=0	70-0=

Table 2-9. Unconstrained Cost Residual Requirement with No Substitution - Part 2 (initial inventory = 10)

.

PART 2 ROMT = LARGEST DAY ROMT = 70

Table 2-10. Unconstrained Cost Residual Requirement with No Substitution - Part 3 (initial inventory = 260)

		 CALCULA ASSUME INIT = 	TE ROMT FOR NEXT MO PREVIOUS (PARTS 1 8 270, NEW PART 2 INI	DST EXPENSIVE PAR 2) RQMTS "BOUGH T = 80	T (PART 3) T" SU THAT NEW PART
DAY	#3 CUM NET DEMAND PART 1 (INIT=270) #1-20 OR (0)	#6 CUM NET DEMAND PART 2 <u>(INIT=80)</u> #4-70 UR (0)	#7 CUM NET DEMAND PART 3 <u>(INIT=260)</u> (FROM 'FULL SUB')	#8 UNALLUCATED ALLOWABLE STUCKUUT #2-#3-#6 OR (0)	<u>DAY RUMT</u> MAX(#7 - #8) UR (0)
1	0	0	0	100	0
2	0	0	0	100	0
3	0	0	0	100	0
4	10	0	0	40	0
5	50	0	40	0	40

2-7

		 CAL ASS INI 	LCULATE RQMT SUME PREVIOUS T = 270, PAR	FOR NEXT MOST EXP (PARTS 1, 2 & 3) T 2 INIT = 80, PA	ENSIVE PART (PA RQMTS "BOUGHT" RT 3 INIT = 500	ART 4) 'SU THAT NEW PART)
	#3 CUM NET DEMAND PART 1	#6 Cum net Demand Part 2	#9 CUM NET DEMAND PART 3®	#10 Cum net Demand Part 4	#11 UNALLUCATED ALLOWABIE	
DAY	<u>(INIT=270)</u> #1-20 OK (0)	<u>(1NIT=80)</u> #4-70 UR (0)	<u>(INIT=300)</u> #7-40 OR (U)	(INIT=30) (FROM FULL SUB)	<u>STUCKOUT</u> #2-#3-#6-#9 UR (0)	<u>DAY RQMT</u> MAX(#10 - #11) OR (U)
1	0	0	0	0	100	0
2	0	0	0	0	100	Û
3	0	0	0	10	100	0
4	10	0	0	20	40	0
5	50	0	0	30	0	60

Table 2-11.	Unconstrained	Cost Residual	Requirement w	ith No	Substitution
	- Part	4 (initial in	ventory = 30)		

Table 2-12. Summary of Unconstrained Cost Residual Requirements

	'FULL ADD-ON	SUB ' Rumt	'NU ADD-ON F	SUB ' Romi
PART 1	20		20	
PART 2	20		70	
PART 3	0		40	
PART 4	0		30	
TOTAL CO	ST 9,000		14,000	
• FROM CI	JM NET DEMAND	COLUMN OF	TABLES 2-4	THRU 2-7

e. Partial-substitution Algorithm Logic. The order of partialsubstitution algorithm operations is described below. They will be illustrated in succeeding paragraphs.

(1) Partition all part types into a full-sub set and a no-sub set as defined in paragraph 2-1a.

(2) Calculate the allowable NMCS aircraft for each day.

(3) For each day:

(a) Generate all possible nonnegative integer combinations (AF, AN) (for full-sub and no-sub, respectively) such that AF + AN = allowable NMCS aircraft for that day.

(b) For each integer combination (AF, AN), compute a basic PARCOM full-sub solution <u>over only the full-sub part</u> set for the scenario through that day, assuming AF allowed NMCS aircraft (awaiting full-sub parts) for that day. Also compute a basic PARCOM no-sub solution <u>over only the no-sub part set</u> for the scenario through that day, assuming AN allowed NMCS aircraft (awaiting no-sub parts) for that day. Calculate the total solution cost for the combination (AF, AN) as the sum of the costs for the full-sub and no-sub solutions described above.

(c) Select the solution for the combination (AF, AN) yielding the minimum total solution cost. This solution consists of the requirements for each part on that day and is called the day requirement. The combination (AF, AN) used in the selected solution then becomes the allowed stockouts used during cumulative (from Day 1) calculations on all succeeding scenario days.

(4) After all days are processed, select the largest (over all scenario days) of the computed day requirements for each part as the overall requirement. The logic for computing a basic PARCOM solution is described in the PARCOM Functional Description.³ The above algorithm tends toward a least cost solution mix (assuming unconstrained funds) for the partial-substitution replacement policy defined by the full-sub/no-sub partition of the part data base.

f. Unconstrained Cost Requirements Under Partial Substitution - Example Conditions

(1) Simplifying Assumptions. The full set of algorithmic calculations was too complex to represent, so for this example only, the following simplifying assumptions were made:

(a) To simplify computation, the combinations (AF, AN) chosen were multiples of 10.

(b) Since, in this example, Day 5 drives (has the largest day requirements for) the solution, the only calculations shown are for Day 5.

(2) Definition of Policy. Part 1 and Part 2 (from Table 2-1) were selected for the full-sub part set and Part 3 and Part 4 for the no-sub part set.

(3) Calculations for Example 1. Table 2-13 shows the algorithm calculations for partial-sub for Day 5. Note that:

(a) The calculation of daily allowable NMCS aircraft used in full sub and no sub also applies here.

(b) For AF = 0 on day 5, the full-sub solution for the full-sub part set is just the largest daily cumulative net demand for each full-sub part. From Table 2-4, this is 70 for Part 1. From Table 2-5, this is 70 for Part 2. These are also the requirements for these parts under an "NMCS = 0" policy in basic PARCOM.

(c) For AF greater than 0, to obtain a full-sub solution based on AF allowed NMCS aircraft for the full-sub parts set, AF x QPA = AF (since QPA = 1 in this example) units are subtracted from each part requirement in the "AF = 0" solution. This is done because each reduction of stock by QPA units creates QPA backorders which, in turn, correspond to one NMCS aircraft.

			ALLOWABL	E NMC	S ACFT = 50		
		AF = ALL	OWABLE NMC	S ACF	T FROM 'FUL	L SUB'	SET
		$A_N = ALL$	OWABLE NMC	S ACF	T FROM 'NO	SUB' SE	T
		'FULL SUB'			'NU SUB'		
		SOLUTION			SOLUTION		COMBINED
COMBINED		PT 1/PT 2			PT 3/PT 4		SOLUTION
<u>SOLUTION #</u>	AE	(\$400/\$50)	COST	AN	(\$40/\$30)	<u>COSI</u>	<u>COST</u>
1	0	70/70	\$31,500	50	0/20	\$600	\$32,100
2	10	60/60	27,000	40	0/30	900	27,900
3	20	50/50	22,500	30	10/30	1,300	23,800
4	30	40/40	18,000	20	20/30	1,700	19,700
5	40	30/30	13,500	10	30/30	2,100	15,600
6	50	20/20	9,000	0	40/30	2,500	11,500
			<u>PT 1</u>	PT 2	<u>PT 3</u> PT 4	<u> </u>	
		MIN COST SU	L = 20	20	40 30		
		(A)	SSUMING DA	Y 5 H	AS THE MAX	RQMT)	

Table 2-13. Unconstrained Cost Residual Requirements Calculations for Day 5 with Partial Substitution - Example 1

(d) For AN = 0 on Day 5, the no-sub solution for the no-sub part set is just the largest daily cumulative net demand for each no-sub part. From Tables 2-6 and 2-7, these are 40 for Part 3 and Part 4. These are also the requirements for these parts under an "NMCS = 0" policy in basic PARCOM.

(e) For AN greater than 0, to obtain a no-sub solution based on AN allowed NMCS aircraft for the no-sub part set, AN units are subtracted from the stock requirement for the most expensive item(s) in the "AN = 0" solution. Each reduction of stock by one unit creates a backorder and corresponds to one NMCS aircraft.

(f) The minimum combined (total) solution cost (\$11,500) is marked in Table 2-13. The combined parts requirement for the associated (AF, AN) combination is the day requirement for Day 5. If (as assumed in this example) Day 5 has the largest day requirement, then that day requirement is also the overall minimum cost solution for our partial-substitution Example 1. From Table 2-12, the resulting solution cost (\$11,500) is between the full-sub solution cost (\$9,000) and the no-sub solution cost (\$14,000).

(4) Calculations for Example 2. The conditions of the previous example are altered slightly in order to illustrate another case. Example 2 is identical to the previous example except for the part unit costs. Table 2-14 shows the new (Example 2) part costs alongside their old (Example 1) values. The following observations apply:

PREVIOUS EXAMPLE	#1 WITH NEW	PART CUSTS AS FULLOWS:	
	PART 1	PART 2 PART 3 PART	4
NEW COST	\$ 40	\$ 50 \$ 400 \$ 30	
OLD COST	\$ 400	\$ 50 \$ 40 \$ 30	

Table 2-14. Part Unit Cost Data for Example 2

(a) Table 2-15 shows the partial-sub solution calculations for Example 2. Note that the Example 1 full-sub and no-sub solutions for (AF, AN) combinations also apply to Example 2. This is true because:

<u>1</u>. Alteration of part unit cost data <u>never</u> changes a full-sub solution.

<u>2</u>. The no-sub solution with new part costs does not change if the cost ordering of no-sub parts is unchanged with the new cost data (since the most expensive items remain the same then). In the given examples, Part 3 is always more expensive than Part 4.

(b) As in Table 2-13, the minimum combined (total) solution cost (\$6,300) is marked in Table 2-15. As before, the combined solution associated with that minimum cost is the day requirement for Day 5 and, by our assumptions, is also the overall minimum cost solution for Example 2. Note that, all else being equal, the relative unit costs of parts drives the partial-sub solution.

Table 2-15.	Unconstrained Cost Residual Requirements Calculations for	or
	Day 5 with Partial Substitution - Example 2	

			ALLO	WABLE	NMCS ACFT	= 50		
		AF = A	LLOWABLE	NMCS	ACFT FRUM	'FULL SU	JB' SET	
		AN = A	LLUWABLE	NMC S	ACET FROM	'NO SUB'	SET	
							021	
		'FULL SUB'			'NO SUB'			
		SOLUTION			SOLUTION		COMBINED	
COMBINED		PT 1/PT 2			PT 3/PT	4	SULUTION	
SOLUTION #	AF	(\$40/\$50)	COST	AN	(\$400/\$30) COST	COST	
1	0	70/70	\$6,300	50	0/20	\$600	\$6,900	
2	10	60/60	5,400	40	0/30	900	6,300	
3	20	50/50	4,500	30	10/30	4,900	9,400	
4	30	40/40	3,600	20	20/30	8,900	12,500	
5	40	30/30	2,700	10	30/50	12,900	15,600	
6	50	20/20	1,800	0	40/30	16,900	18,700	
			P	T 1	PT 2 PT 3	PT 4		
		MIN COST	S01 =	60	60 0	30		
			(ASS0M1N	G DAY	5 HAS THE	MAX ROMT)	
					2 mile mile	THE THE PARTY		

2-3. APPLICATION TO FULL-SCALE DATA BASE

a. Background. The previous section treats relatively simple stylized examples of little practical interest. In this section, requirement results are presented for extended PARCOM applied to the AH-1S helicopter parts data base and scenario used in the CAA study reports for the Aircraft Spare Stockage Methodology Study and the MAX FLY Study.⁴ The extended PARCOM case will be denoted as the MAX FLY example. The associated parts data base has 334 different part types tagged as essential to aircraft operation. That data base was applied in a 120-day European scenario.

b. Partial Substitution Policy. A partial-substitution policy in extended PARCOM is defined in terms of the part types (in the parts data base) which comprise the full-sub parts set (see para 2-1). The full-sub parts of the MAX FLY example were defined as all those part types with either a not repairable this station (NRTS) rate exceeding 50 percent or with a retail repair time (as specified in the data base) of at least 30 days. The resulting full-sub part set employed by extended PARCOM contains 102 part types. All other part types are in the no-sub parts set. Again, a full-substitution policy is just a special case of partial substitution policy is a special case in which no part type is in the full-sub set.

c. Comparative Results. Table 2-16 summarizes the comparative residual (add-on to current inventory) requirement results, by replacement policy, for the MAX FLY data base and scenario. The partial-substitution policy represented therein is the one defined above. The relatively small difference between the partial-substitution and full-substitution requirement costs is primarily due to the dominance of the requirement costs for a single part type, the stability control amplifier, in all three policy cases. The full-substitution cost and the no-substitution cost are lower and upper bounds, respectively, on all partial-substitution policies. The partial-substitution policy applied here is just one of many potential policies. If new partial-substitution policies are defined by transferring some no-sub part types to the full-sub part set, then the associated requirements costs will decrease and will approach the full-substitution policy cost. Conversely, if policies are defined by transferring some full-sub part types into the no-sub set, then the associated requirements costs will increase and approach the no-substitution policy cost. The size of the change in requirements cost associated with an altered full-sub part set (and hence a different partial-substitution policy) depends on which parts are added to and/or removed from the base full-sub set. Further sensitivity studies, not performed here, would be needed to explore the comparative and marginal effects of variation in the partial-substitution policy employed.

Table 2-16.	Add-on Requirements	Costs by	Policy	- MAX FLY	Example
-------------	---------------------	----------	--------	-----------	---------

Policy	Add-on	Number of part	Largest part
	cost, \$M	types w/add-on	rqmt (% of total)
Full substitution	20	6	99
Partial substitution	21	60	94
No substitution	43	99	72

2-4. WORKAROUND - AN APPROACH THAT FAILED

a. Background. At the start of OPTP, when it was first determined that partial substitution should be investigated, the prospects for successfully designing appropriate partial-substitution logic for PARCOM and the ease of integrating it into the model were unknown. In view of those uncertainties, it seemed desirable to seek some simple way of working around the limitations of the version of the model in use at that time by developing some kind of input or run modifications that would permit PARCOM to effectively represent partial substitution without changes having to be made in the program code. An approach that seemed feasible at the time is described below.

b. Approach. First, an unconstrained cost residual requirements case is run with a full-sub parts replacement policy. Next, the same case is run with a no-sub policy. Relative to all possible part replacement policies, the former generates the smallest number of required parts and part types and the latter the largest. In order to represent a partial-sub case, one assumes that some parts from the no-sub requirements list are substitutable and would not be required in a partial-sub run if they are so designated. The appropriate substitutable parts are those showing up as required in the full-sub run. For the partial-sub run, then, two sets of parts are established. One set consists of those part types designated as required in a full-sub run, plus those additional part types designated as substitutable (which are associated with the "holes" in the NMCS aircraft generated in the full-sub run). The second set consists of the remaining nonsubstitutable part types. The workaround solution to a partial-sub requirements run is just the original full-sub solution, plus the no-sub, NMCS = 0 solution for the set consisting of the remaining, nonsubstitutable parts. NMCS = 0 is appropriate for this set, since all the allowable NMCS aircraft are assumed "locked up" supplying parts to the full-sub set.

c. Results. The above approach was tested with the example cases of the previous section. For Example 1, the extended PARCOM and workaround solutions are the same--the set of required parts costing \$11,500 in each case. For Example 2, however, the workaround solution is three times as expensive as the PARCOM direct modeling solution--\$18,700 versus \$6,300-- thus proving that the workaround approach does not always provide the right answer. The difference is due to the assignment, in Example 2, of greatest cost to Part 3, one of the nonsubstitutable set. It appears that whenever one of the parts from this set is the highest cost part, the workaround solution may not be optimum, depending also on failure rates and other factors. The workaround approach to partial substitution was therefore discontinued, especially since appropriate partial substitution logic for PARCOM had, meanwhile, been accomplished.

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CHAPTER 3

CAPABILITY ASSESSMENT WITH PARTIAL SUBSTITUTION

3-1. BACKGROUND. After each unconstrained cost solution mix is computed, PARCOM generates a record of daily and average fleet operational capability achievable by stocking each computed requirement. In particular, these records include achieved daily and average aircraft availability, achieved program flying hours, and achieved flying hours per available aircraft per day. In computing these outputs, the new initial inventory is assumed to be the sum of the computed requirement and the original initial inventory.

3-2. CHAPTER ORGANIZATION. Subsequent paragraphs first illustrate the basic PARCOM capability assessment under full substitution and under no substitution for the unconstrained cost requirements of the example case defined in the previous chapter (Tables 2-1 and 2-2). The extension to partial substitution is then shown for its example cases (defined in Chapter 2). Lastly, assessment of current inventory is portrayed.

3-3. ASSESSMENT WITH FULL SUBSTITUTION. Tables 3-1 and 3-2 show the basic PARCOM capability assessment calculations, under full substitution, of the expected effects of stocking the requirements computed in Tables 2-4 through 2-7, Chapter 2. Cumulative net demand for each part type is based on initial inventories being set to include the computed requirements. NMCS aircraft for each day are set equal to the largest of the "cumulative net demand/QPA" entries for the day. "Surviving aircraft" are from the "cum acft surv" column of Table 2-2. Aircraft availability is 1 minus the quotient of NMCS aircraft and surviving aircraft. Flying hours per (available) aircraft per day are calculated by dividing the program flying hours for each day (see Table 2-2) by the number of available aircraft on that day. Average availability is constructed by weighting daily availabilities by the daily surviving aircraft. Average flying hours per (available) aircraft per day is weighted by the available aircraft on each day.

	#1	#2	#3	#4	#5
	CUM NET	CUM NET	CUM NET	CUM NET	
	DEMAND/QPA	DEMAND/QPA	DEMAND/QPA	DEMAND/QPA	
	PART 1	PART 2	PART 3	PART 4	NMC S
AY	<u>(INIT=270)</u>	(INIT=30)	(INIT=260)	(INIT=30)	ACET
	(ORIG*-20)/QPA	(URIG*-20)/QPA	(ORIG*-0)/QPA	(ORIG*-0)/QPA	MAX (#1,#2,#3,#4
	(OR O)	(OR 0)	(UR 0)	(OR 0)	
1	0	0	0	0	0
2	0	0	0	0	0
2	U	0	U	U	U
3	0	20	0	10	20
4	10	40	0	20	40
5	50	50	40	30	50

Table 3-1. Capability Assessment for Unconstrained Cost Residual Requirement with Full Substitution

Table 3-2. Capability Assessment for Unconstrained Cost Residual Requirement with Full Substitution (continued)

DAY	#5 NMCS <u>ACFI</u>	#6 SURVIVING <u>ACFI</u> DATA	#7 ACFT <u>AVAILABILITY</u> 1 #5/#6	FLYING HUURS <u>/ACFI/DAY</u> FHP/(#6 x #7)
1	0	150	1.00	3.3
2	0	200	1.00 ,	5.0
3	20	200	• 90	5.6
4	40	200	-80	9.4
5	50	200	.75	10.0
AVG AV	AIL = .88			
AVG FH.	/ACFT/UAY = 6.5			

3-4. ASSESSMENT WITH NO SUBSTITUTION. Tables 3-3 and 3-4 show the basic PARCOM capability assessment calculations, under no substitution, of the expected effects of stocking the requirements computed in Tables 2-8

through 2-11. Cumulative net demand for each part type is based on initial inventories being set to the computed requirements. Under a no-substitution policy, NMCS aircraft for each day are equal to the sum of the cumulative net demand entries for that day. Surviving aircraft are from Table 2-2. Other calculations are analogous to those for the full-substitution case.

	RI	ESIDUAL ROMT (20,70,	40,30) IS ADDED	TO ORIGINAL INIT	(250,10,260,30)
	#1	#2	#3	#4	#5
	CUM NET	CUM NET	CUM NET	CUM NET	
	DEMAND	DEMAND	DEMAND	DEMAND	
	PART 1	PART 2	PART 3	PART 4	NMC S
DAY	<u>(INIT=270)</u>	<u>(INIT=80)</u>	(INIT=300)	(INIT=60)	ACFT
	(ORIG*-20)	(ORIG*-70)	(OR16*-40)	(ORIG*-30)	SUM OF #1 - #4
	(UR O)	(OR O)	(OR O)	(OR O)	
1	0	0	0	0	0
2	0	0	0	0	0
3	0	0	0	0	0
4	10	0	0	0	10
5	50	0	0	0	50
• 0	RIGINAL CUM N	ET DEMAND BASED ON O	RIGINAL INIT		

Table 3-3. Capability Assessment for Unconstrained Cost Residual Requirement with No Substitution

Table 3-4. Capability Assessment for Unconstrained Cost Residual Requirement with No Substitution (continued)

DAY	#5 NMCS <u>ACEI</u>	#6 SURVIVING <u>ACFT</u> DATA	#7 ACFT <u>AVAILABILITY</u> 1 #5/#6	FLYING HOURS <u>/ACET/DAY</u> FHP/(#6 x #7)
1	0	150	1.00	3.3
2	0	200	1.00	5.0
3	0	200	1.00	5.0
4	10	200	.95	7.9
5	50	200	.75	10.0 .
AVG AVA	AIL = .94			
AVG FH	ACFT/DAY = 6.2			

3-5. ASSESSMENT WITH PARTIAL SUBSTITUTION

a. Example 1. Tables 3-5 and 3-6 show the extended PARCOM capability assessment calculations, under the partial-substitution policy of Chapter 2 (Part 1 and Part 2 are the full-sub set), of the effects of stocking the Example 1 requirements computed in Table 2-13. Each day consists of a full-sub assessment phase and a no-sub assessment phase. Each full-sub phase is equivalent to a basic PARCOM full-sub assessment of NMCS aircraft with only the full-sub part set considered. The resulting NMCS aircraft for the day are computed as in Table 3-1. The no-sub phase is equivalent to a basic PARCOM no-sub assessment of NMCS aircraft with only the no-sub part set considered. Resulting NMCS aircraft for the day are computed as in Table 3-3. Under our definition of partial substitution, each NMCS aircraft is "down" due to either at least one needed full-sub part or for a single needed no-sub part, but not to a needed combination of the two types. Therefore, the order of performing the phases is irrelevant. On each day, after the two NMCS aircraft calculation phases are completed, the sum of the two results yields the total NMCS aircraft for the day. Other calculations on Table 3-6 are exactly analogous to those applied by basic PARCOM in Tables 3-2 and 3-4.

		RESIDUAL ROM	T (20,20,40,30) I	S ADDED TO UR	IGINAL INIT (250,10,260,30)
		#1	#2	#3	#4	#5
		CUM NET	CUM NET	CUM NET	CUM NET	
		DEMANU/QPA	DEMAND/QPA	DEMANU	DEMAND	
		PART 1	PART 2	PART 3	PART 4	NMC S
DAY	PHASE*	(1NIT=270)	(INIT=30)	(INIT=300)	(INIT=60)	ACFT
		(ORIG**-20)/QPA	(ORIG**-20)/QPA	(OR16**-40)	(OR16**-30)	MAX (#1,#2) (FS PHASE)
		(OR O)	(UK 0)	(OR O)	(OR 0)	#3+ #4 (NS PHASE)
1	FS	0	0			0
	NS			0	0	0
2	FS	0	0			0
	NS			0	0	0
3	FS	0	20			20
	NS			0	0	0
4	FS	10	40			40
	NS			0	0	0
5	FS	50	50			50
	NS			0	0	0
•	FS = 'FL NS = 'NU	ILL SUB' PHASE (PR SUB' PHASE (PRUC	OCESSES 'FULL SUB ESSES 'NO SUB' PA	' PART SET (P RT SET (PARTS	ARTS 1 & 2)) 3 & 4))	
••	ORIGINAL	. CUM NET DEMAND B	ASED ON ORIGINAL	INIT		

Table 3-5. Capability Assessment for Unconstrained Cost Residual Requirement with Partial Substitution - Example 1

	#6 Total	#7 _	#8	
DAY	NMCS <u>ACFT</u> #5(FS) + #5 (NS)	SURVIVING ACET DATA	ACFT AVAILABILITY 1 #6/#7	FLYING HUURS /ACFT/DAY FHP/(#7 x #8)
1	0	150	1.00	3.3
2	0	200	1.00	5.0
3	20	200	•90	5.6
4	40	200	•80	9.4
5	50	200	•75	10.0
AVG	AVAIL = .88			
AVG	FH/ACFT/DAY = 6.5			

Table 3-6. Capability Assessment for Unconstrained Cost Residual

b. Example 2. Tables 3-7 and 3-8 show the extended PARCOM capability assessment calculations, under the partial-substitution policy of Chapter 2, of the effects of stocking the Example 2 requirements computed in Table 2-15. Calculations are exactly analogous to those of Tables 3-5 and 3-6.

Table 3-7. Capability Assessment for Unconstrained Cost Residual Requirement with Partial Substitution - Example 2

		RESIDUAL ROMT	(60,60,0,30) IS AI	DED TO URIGIN	AL INIT (250,	,10,260,30)	
		#1	#2	#3	#4	#5	
		CUM NET	CUM NET	CUM NET	CUM NET		
		DEMAND/QPA	DEMAND/QPA	DEMAND	DEMAND		
		PART 1	PART 2	PART 3	PART 4	NMC S	
DAY	PHASE*	(INIT=310)	(INIT=70)	(INIT=260)	(1N1T=60)	ACFT	
		(ORIG**-60)/QP/	A (ORIG**-60)/QPA	(OR16**-0)	(OR16**-30)	MAX (#1,#2)	(FS PHASE)
		(OK O)	(UR 0)	(OR 0)	(UK 0)	#3+ #4	(NS PHASE)
1	FS	0	0			0	
	NS			0	0	0	
2	FS	0	0			0	
2	1.5	U	U	0	2	0	
	ND			U	U	U	
3	FS	0	0			0	
	NS		••	0	0	0	
4	FS	0	0			0	
	NS			0	0	0	
	NO			0	0	0	
5	FS	70-60=10	0			10	
	NS	••		40	0	40	
• F	S = 'FUL	L SUB' PHASE (PI	ROCESSES 'FULL SUE	" PART SET (P	ARTS 1 & 2))		
N	S = 'NU	SUB' PHASE (PRO	CESSES 'NO SUB' PA	ART SET (PARTS	3 8 4))		
•• 0	PIGINAL	CHM NET DEMAND	RACED ON OPTGINAL	INIT			

3-5
	TOTAL			
	NMCS	SURVIVING	ACFT	FLYING HOURS
DAY	ACFT	ACFT	AVAILABILITY	/ACFT/DAY
	#5(FS) + #6 (NS)	DATA	1 #6/#7	FHP/(#7 x #8)
1	0	150	1.00	3.3
2	0	200	1.00	5.0
3	0	200	1.00	5.0
4	0	200	1.00	7.5
5	10+40=50	200	.75	10.0
AMC A	VAL = 05			
AVU A	VAIL 35			

Table 3-8. Capability Assessment for Unconstrained Cost Residual Requirement with Partial Substitution - Example 2 (continued)

3-6. ASSESSMENT OF CURRENT INVENTORY WITH PARTIAL SUBSTITUTION

a. Logic. By current inventory is meant any user-specified inventory. This is in contrast to the "required inventory" as assessed above. The basic logic of assessment of current inventory in extended PARCOM is the same as in basic PARCOM. With unconstrained costs, net demand was based on the entire planned flying hour program being flown. For a current inventory mix, some unknown (at first) number of hours will be flown. That number must initially be estimated and an iterative approach applied to determine NMCS aircraft, availability, and achievable program flying hours. For each day, therefore, a starting estimate of flying hours flown is made. The starting (first day's) estimate is the program flying hours. Then, net demand, as based on the estimated flying hours, is computed, followed by implied NMCS aircraft (generated by the estimated flying hours), achievable flying hours, and flying hours per available aircraft. The achievable flying hours are compared with the estimated flying hours flown. If, based on input thresholds, they are close enough, the iterations stop. If not, the calculations are repeated based on a new starting estimate of flying hours equal to the average of the estimated and the achieved flying hours. After iterations for a day are completed, the available aircraft for the day and their flying hour potential are calculated based on the last calculation of NMCS aircraft and on the maximum flying hour potential per aircraft per day (an input). Processing for the next day uses a starting estimate of flying hours based on the achieved flying hours of the previous day.

b. Example. Tables 3-9 through 3-11 show the extended PARCOM current inventory capability assessment calculations, through 4 days, for the example of Tables 2-1 and 2-2. For example purposes, iterations are limited to two. Calculation of daily NMCS aircraft is done in two phases, as before, but cumulative net demand is based on the current inventory and on the estimated flying hours for the iteration. The NMCS aircraft for the last iteration of each day become the basis of final daily calculations. In column 7 calculations, surviving aircraft are from Table 2-2, while NMCS aircraft are from column 6. In column 8, achieved flying hours are capped at the daily program flying hour objective. Column 9 shows the calculation of closeness thresholds for estimated versus achieved flying hours. As in basic PARCOM, the model user sets the limits on iterations and closeness thresholds.

				INVENTORY	= (250,10,260,30)	ITERATION L	IMIT = 2	
DAT	ITER- ATION	PHASE •	ÈST FLY <u>HRS</u> **	#1 Cum NET DEMAND*** PART 1 (INIT=250)	#2 COM NET DEMAND*** PART 2 (INIT=10)	#3 COM NET DEMAND*** PART 3 (INIT=260)	#4 CUM NET DEMAND*** PART 4 (INIT=30)	#5 NMCS ACFT MAX (#1,#2) QR (#3 + #4)
1	$\frac{1}{1}$	FSNS	500 500		0	 0		0 0
2	$\frac{1}{1}$	FS NS	1000 1000	0	20	0		20 0
3	$\frac{1}{1}$	FS NS	1000 1000	0	40	0	10	40 10
4	$\frac{1}{1}$	FSNS	1500 1500	30	60	0	20	60 20
	22	FSNS	1350 1350	18	57	0	17	57 17
	• FS •• = ••• CA	= 'FULL FHP UN I LCULATED	. SUB' F TERATIO AS COM	PHASE; NS = 'NU DN 1; = (EFH + . N FAILORES - COL	SUB' PHASE AFH)/2 ON ITERATION M RETORNS - INIT IN	2 VENTURY		

Table 3-9. Capability Assessment of Current Inventory with Partial Substitution

3-7

		WILL FALLA	JUDSTITU	cron (concinued)	
		#6 Tutal	#7	#8	#9
•	ITER-	NMCS	AVAIL	ACHIEVED	(EFH - AFH)/
DAY	ATION	ACFT	ACFT	FLYING HKS	(AVG DAY FHP)
		#5(FS) + #5 (NS)	SURV-	MIN(#7 X MFHAD*)	
			NMCS	UR (FHP)	
1	1	0	150	500	0
2	1	20	180	1000	0
3	1	50	150	1000	0
4	1	80	120	1200	•27
	2	74	126	1260	.08
* M	AXIMUM FL	YING HOURS PER ACFT	PER DAY		

Table 3-10. Capability Assessment of Current Inventory with Partial Substitution (continued)

Table 3-11.	Capability Assessment o	of Current Inventory with
	Partial Substitution ((continued)

	#10		FRAC
	SURVIVING	ACFT	FLYING PGM
DAY	ACFT	AVAIL	ACHIEVED
	DATA	#7/#10	#8/FHP
1	150	1.00	1.00
2	200	•90	1.00
3	200	•75	1.00
4*	200	,63	.84
* LAST	ITERATION'S	VALUES	

c. Full-scale Data Base Application. Figure 3-1 shows comparative (by policy) capability assessement of current inventory, in terms of fraction of daily flying program achieved, for the MAX FLY example case of paragraph 2-3. While the partial-substitution policy has almost one-third of the data base parts in the full-sub set, there is only a small difference between program flying hour achievement under partial substitution and under no substitution. Part of the reason is that the mix of parts comprising the full-sub set under the chosen partial-substitution policy is probably not the best one in terms of maximizing fleet capability. Apparently, the criteria defining the chosen partial-substitution, full-substitution set (NRTS > .50 or retail repair time > 30) do not correlate closely with performance. That policy does have a plausible aspect in that parts which are repaired at depot and/or which have a long repair cycle time appear to be more likely candidates for substitution. However, items with high failure rates may be more appropriate as members of the full-sub set. Preliminary testing indicates that this may be so. In any case, Figure 3-1 suggests that use of partial substitution may not always be justified by the returns in terms of improved flying hour productivity.



Figure 3-1. Capability Assessment of Current Inventory -MAX FLY Example Case

CHAPTER 4

DISTRIBUTING PARTS OVER TIME

4-1. BACKGROUND. The basic PARCOM Model assumes that all spare assets are front-loaded, i.e., that they are all available at retail on Day 1 of the scenario. The Overview Model (evaluated in the Aircraft Spare Stockage Methodology Study) allows the user to specify a phasing-in of parts (into theater) over time. Such phasing-in is more representative of reality since it reflects movement of unit ASL/PLLs and transit of depot stocks. The extended PARCOM was reconfigured to allow initial stock to be received in theater according to a specified planning schedule. The planning schedule is assumed operative; disruptions in the schedule, due to attrition of resupply lines and facilities, is not simulated.

4-2. LOGIC. Extended PARCOM distributes parts over intervals of 5 days rather than over individual days, as in Overview. All parts due to be received during a given 5-day interval are distributed uniformly throughout that interval. An exception is Day 1 of the scenario. All parts due in (or in place) on Day 1 are treated as received at the beginning of Day 1. The categories of parts treated are as follows:

a. Depot Serviceables. These consist of serviceable parts located at depot at the start of the scenario. For each part, the initial stock of depot serviceables is entered in the part data base input. The scenario input specifies a depot lag, L, and a depot distribution time, D, applicable to all parts, such that, for each part, the initial stock of depot serviceables is distributed (received at retail) uniformly between Day (L + 1) and Day (L + D).

b. Depot Unserviceables. These consist of unserviceable parts located at depot at the start of the scenario. They are at various stages of the depot repair process and, after repair, are to be shipped to retail. Since a part may be at any stage of its repair cycle, distribution of uncondemned depot unserviceables for each part is assumed uniform over an interval equal to the depot repair time (DRT) for the part, with the first receipt (at retail) after a lag equal to the order/ship time (OST) for the part. For each part, the initial stock of depot unserviceables, the depot condemnation rate (DC), the OST, and the depot repair time are input in the part data base. Letting A = number of depot unserviceables, the extended PARCOM distributes (1-DC) x A parts at retail between Day (OST + 1) and Day (OST + DRT).

c. War Reserve Serviceables. These consist of serviceable parts in the war reserve located at retail. For each part, the amount of the serviceable war reserve is input in the parts data base. The entire stock is treated as available at retail from the scenario start (Day 1).

d. War Reserve Unserviceables. These consist of unserviceable war reserve parts located at retail at the start of the scenario. Some of these will be condemned. Others will be sent to depot for repairs. Others are in various stages of repair at retail. The distribution of these parts is as follows:

(1) Items repairable at retail - for each part, let NRTS = the NRTS fraction, BR = the retail repair time, BC = retail condemnation rate, and A = number of war reserve unserviceables. Then extended PARCOM distributes at the theater $(1-NRTS) \times A \times (1-BC)$ parts repaired at retail between Day 1 and Day BR. All of these factors are input in the parts data base.

(2) Items not repairable at retail - for each part, let NRTS = the NRTS fraction, DR = the depot repair time, DC = depot condemnation rate, OST = the order/ship time, and A = number of war reserve unserviceables. Then extended PARCOM returns to the theater (NRTS) x A x (1-DC) parts repaired at depot between Day (2 x OST + 1) and Day (2 x OST + DR).

e. ASL/PLL Deployments. For each part, the extended PARCOM parts data base inputs on Day 1 the total in-place ASL/PLL parts. In addition, total ASL/PLL parts deployed after Day 1 are input for successive 5-day intervals of the scenario.

4-3. IMPACT. The distribution of parts over time, as opposed to front loading of stocks, has no effect on PARCOM results if all initial assets reach retail before they are required (as replacements). An ideally efficient stockage and transportation system will achieve this. Parts distribution over time may effect an increase in requirements, relative to front loading, if initial assets are sufficiently delayed so that they do not arrive in retail before all retail stocks are drawn down. In effect, such delayed assets may have their usefulness negated because they are in the wrong place at the wrong time. Similarly, the effect of such delays on capability assessment of current inventory may be a decrease in the period over which the flying program can be continuously sustained.

4-4. EXAMPLE RESULTS. A comparative example is presented of the effects of part maldistribution in the full-substitution demonstration example of Chapter 2, which assumed front loaded parts. The parts data of Table 2-1 are used, except that Part 1 initial stock is distributed over time as specified in Table 4-1. Since just Part 1 data is altered, only the full-substitution requirement for that part is recalculated by revising Table 2-4 in accordance with the parts distribution. Table 4-2 shows the revised calculations. The basic change is in column number 3, in which INIT (front loaded initial stock) of Table 2-4 is replaced by STK (cumulative stock distributed) from Table 4-1. The net result is that cumulative net demand through Day 4 is larger under parts distribution. The overall requirement is larger (70, versus 20 for the front loaded case) because the parts deployment is badly timed. On Days 2 through 4, net demands exist while initial assets are unable to fill them, due to distribution delay.

4-2

ALL INITIA	L STOCKS UNCHANGED EXCEPT
- PART 1	DISTRIBUTIUN
	CUM STUCK
DAY	DISTRIBUTED
1	40
2	80
3	120
4	160
5	250

Table 4-1. Example - Part 1 Stock Distribution Over Time

Table 4-2.	Unconstrained Cost Residual Requirement with Full Substitution	-
	Part 1 (initial stock distributed over time)	

		PAR	T 1 CALCULATIONS		
DAY	#1 CUMULATIVE FAILURES	#2 CUM RET <u>REPAIRS</u>	#3 CUMULATIVE <u>NET DEMAND</u> MAX [#1-(#2+STK*)] OR (0)	#4 ALLOWABLE STUCKOUTS	DAY <u>RQMTS</u> MAX (#3-#4) UR (0)
1	40	0	40 - 0 - 40 = 0	· 100	0
2	120	0	120 - 0 - 80 = 40	100	0
3	200	0	200-0-120 = 80	100	0
4	320	40	320-40-160 = 120	50	70
5	440	120	440-120-250 = 70	50	20

PART 1 OVERALL ROMT = LARGEST DAY ROMT = 70

• CUMULATIVE STOCK DISTRIBUTED FROM TABLE 4-1

CHAPTER 5

CONSTRAINED COST REQUIREMENTS

5-1. BACKGROUND. While the unconstrained cost solution is the one that best meets the flying program, a full requirements buy may not be affordable if funds are limited. With constrained costs, a user wishes to apply limited funds to buy a cost effective slice of the full requirements. The basic PARCOM only treated the constrained-cost case for a no substitution policy. Neither full substitution nor partial substitution were addressed. The extended PARCOM incorporates a method for deriving cost effective constrained cost requirements under partial substitution. For a no-substitution policy, the extended PARCOM constrained cost algorithm yields the same solution as the basic PARCOM constrained cost algorithm.

CONSTRAINED COST NO-SUBSTITUTION REQUIREMENT IN BASIC PARCOM. This 5-2. algorithm is covered in the PARCOM Functional Description. To summarize, after the unconstrained cost, no-substitution requirements are computed, they become the basis for the constrained cost no-substitution solution. A cost limit on spares is input along with the other scenario and objective data. A constrained cost, no-substitution parts mix can be constructed by the simulated purchase, in order of increasing part unit cost, of the part requirements of the unconstrained cost solution until the money is exhausted. That would entail the procurement of the largest number of total parts from the unconstrained cost solution. However, another characteristic of such a constrained cost parts mix is that it is the mix which has the fewest unbought (hence, unstocked) items from the unconstrained cost solution. The PARCOM algorithm arrives at its solution by calculating unbought items. Initially, it spends the full cost of the unconstrained cost requirements mix, assuming it to be the constrained cost solution. Subsequently PARCOM selects the fewest number of items to remove from that solution until the remaining parts mix is priced at the input cost limit. Because the programed algorithm solves by unbuying items rather than buying them, parts are processed in decreasing order of part unit cost. Under a policy of no substitution each unbought item (regardless of part type) creates an NMCS aircraft. Therefore, our constrained cost, no-substitution solution mix minimizes the instances of NMCS created by the constrained funds. The solution tends, heuristically, toward the achievement of maximum cumulative flying hours.

5-3. APPROACH IN EXTENDED PARCOM. First, a method for treating full substitution was devised. The basic PARCOM constrained cost algorithm for no substitution was retained and combined with the full-substitution algorithm to yield a composite algorithm applicable to all partial substitution cases. However, since this algorithm is not known to be the best in all cases, its solution is compared, in terms of resulting program flying hour productivity, with a solution derived by another algorithm. The solution yielding the most program flying hours is selected. Herein, we denote these two algorithms as constrained cost algorithm one, and constrained cost algorithm

two respectively. Since both are based on the approach for a full-substitution case, that case will be discussed first, followed by its adaptation to the two constrained cost algorithms.

5-4. CONSTRAINED COST WITH FULL SUBSTITUTION. For this case, the constrained cost solution is equivalent to using maximum consecutive days of flying hour program achievement as an objective. This algorithm is described in the PARCOM Functional Description. The nature of full substitution is such that the solution yielding the maximum consecutive days of flying program sustainability for fixed funds will also be the solution yielding maximum total program flying hour productivity. As was shown in the Aircraft Spare Stockage Methodology Study Report, this was not the case with a no-substitution policy. The algorithm for obtaining such a maximum sustainability solution also is described in the basic PARCOM Functional Description. Solution generation in extended PARCOM is automatic for full substitution.

5-5. CONSTRAINED COST ALGORITHM 1. After the unconstrained cost partialsubstitution requirements are computed, they become the basis for a constrained cost solution as follows:

a. The no-substitution constrained cost algorithm described in paragraph 5-2 is applied to yield the portion of the unconstrained cost requirement for the no-sub part set which yields the most cost-effective mix of no-sub parts priced at (or below) the input cost limit. If the input cost limit is less than or equal to the cost of the unconstrained cost no-sub requirement, then the algorithm solution for the no-sub set is the overall solution and the algorithm terminates. However, if the input cost limit exceeds the unconstrained cost no-sub requirement, then that entire requirement is assumed bought and the input cost limit is adjusted by subtracting the cost of the entire no-sub requirement from it. The second phase of the algorithm (below) is then applied with this adjusted cost limit.

b. During the second phase of the algorithm, a version of the fullsubstitution constrained cost algorithm described in paragraph 5-4 is applied to the full-sub part set using the adjusted cost limit as follows:

(1) During the solution of the unconstrained cost case, the model stores, for each day, the cumulative total cost of all the full-sub parts in the partial-substitution unconstrained cost requirement for the scenario truncated at that day. The model determines D, the last (latest) day for which the associated cumulative requirement cost of the full-sub set is less than or equal to the adjusted cost limit.

(2) Next, the model generates an unconstrained cost partial-substitution solution for the scenario truncated at that day. The full-sub parts required in that solution, when combined with the no-sub requirement which was bought in the first phase, comprise the overall algorithm solution. There is no guarantee that the above solution is optimum, but it does combine the two algorithms discussed earlier.

5-2

5-6. CONSTRAINED COST ALGORITHM 2. This algorithm is a version of the maximum sustainability solution described in paragraph 5-4. It will generate a solution yielding the maximum consecutive days of program flying hour achievement. However, the resulting solution may not yield maximum total program flying hours achievable. The algorithm is:

a. During solution of the unconstrained cost case, the model stores, for each day, the cumulative total cost of all parts in the partialsubstitution unconstrained cost requirement for the scenario truncated at that day. The model then determines D, the latest day for which the associated cumulative total requirement cost is less than or equal to the cost limit.

b. Next, the model generates an unconstrained cost partial-substitution solution for the scenario truncated at that day. The resulting solution mix is the overall algorithm solution.

5-7. SOLUTION SELECTION. The preferred solution mix, of those generated by the two algorithms, is the one which yields the maximum program flying hour productivity in the scenario. The model therefore does two separate current inventory capability assessments of the current inventories based on the two constrained cost algorithm solutions being bought and stocked. The add-on solution requirement is assumed to be added to the war reserve. The final constrained cost solution is the one (of the two generated) for which the associated capability assessment yields the larger value for average fraction total flying hour program achieved.

5-8. SAMPLE RESULTS. To illustrate the algorithm described above, the extended PARCOM was applied, in a constrained cost mode, to the partialsubstitution MAX FLY example of paragraph 2-3. Table 5-1 summarizes requirement costs with an unconstrained budget. Total cost is the sum of the cost of full-sub parts and of no-sub parts. Three cost limits, as shown in Tables 5-2 and 5-3 were applied. Table 5-2 shows the comparative results, in terms of flying hour productivity, of the two constrained cost algorithms described previously. Notice that the solution selection, using the preferred algorithm, is based on algorithm 2 in one case and algorithm 1 in two cases. Table 5-3 shows the composition of costs of the constrained cost requirement. In this case, the no-sub parts seem to be preferred by the extended PARCOM algorithm. For the example cost limit (\$.2M) which is less than the total cost of no-sub parts in the unconstrained cost requirement (\$1.1M in Table 5-1), only no-sub parts are bought. For the example cost limits (\$2M, \$3M) which exceed the total cost of no-sub parts with unconstrained budget, all of the no-sub parts in the unconstrained budget requirement are bought. Algorithm 1 always prefers no-sub part purchases. However, algorithm 2 may buy a mix of both.

Tatal anat (fW)	Cost (\$M) by part set		
lotal cost (\$M)	Full sub	No sub	
21.0	19.9	1.1	

Table 5-1.	Add-on Requirements Costs - Unconstrained Budget w	ith
	Partial Substitution - MAX FLY Example	

Table 5-2. Comparison of Constrained Cost Algorithms - Add-on Requirements - Partial Substitution - MAX FLY Example

Cost limit (fW)	Fraction flying	Ductorned	
Cost Timit (am)	Algorithm 1	Algorithm 2	algorithm
0.2 2.0 3.0	.49 .81 .83	.54 .58 .62	2 1 1

Table 5-3. Add-on Requirements Costs - Constrained Budget with Partial Substitution - MAX FLY Example

C+ 1:-:+ (fw)	Solution cost (\$M) by part set		
	Full-sub parts	No-sub parts	
0.2	0.0	0.2	
2.0 3.0	0.9 1.9	1.1 1.1	

CHAPTER 6

OBSERVATIONS

6-1. PARTIAL-SUBSTITUTION REQUIREMENTS. Extended PARCOM is restricted to partial substitution policies in which all part types are partitioned by the model user into a full-sub set, within which all parts are substitutable, and a no-sub set, within which no parts are substitutable. However, considerable flexibility is allowed by such policies. Iterative, automated application of basic PARCOM logic enables calculation of least-cost requirements solutions under partial substitution and a no-sub set within which no parts are substitutable. Example application showed extended PARCOM to give plausible results with partial substitution costs between (low) costs under full substitution and (high) costs under no substitution.

6-2. PARTIAL-SUBSTITUTION CAPABILITY ASSESSMENT. Extended PARCOM can evaluate fleet capability (availability, fraction flying program achieved) for an input-specified initial spares inventory or for a spares inventory reflecting a PARCOM requirements solution being stocked. Example applications showed plausible results with fleet capability under partial substitution between (low) capability under no substitution and (high) capability under full substitution.

6-3. PARTS DISTRIBUTED OVER TIME. Extended PARCOM allows initial spare stocks to be deployed to retail in 5-day intervals, according to user input. Example applications showed plausible results with spare requirements increasing if initial stocks are withheld so long that they are unavailable when needed at retail.

6-4. PARTIAL SUBSTITUTION WITH CONSTRAINED COST. A constrained cost solution algorithm for the full substitution case was developed. This was combined with the basic PARCOM solution algorithm for the no-substitution case to yield a composite algorithm for treating constrained cost under partial substitution in extended PARCOM. However, since the algorithm does not always give the best solution (i.e., the one yielding maximum achievable program flying hours with the constrained funds), a second algorithm was also devised. Extended PARCOM applies both algorithms and chooses the solution mix from the one yielding higher flying productivity. Example results appeared plausible.

APPENDIX A

EXTENDED PARCOM - INPUT SUMMARY

A-1. PARTS DATA BASE INPUT. The major portion, in terms of quantity of records, of the extended PARCOM input data is the parts data base. The elements shown in Table A-1 must be input for each part type used.

Table A-1. Data Elements for Each Part Type in the Parts Data Base

1. National stock number (NSN) 2. Unit cost 3. Retail repair time 4. Depot repair time Order and ship time 5. 6. Failure rate Retail NRTS rate 7. 8. Retail condemnation percentage 9. Depot condemnation percentage 10. Item essentiality code 11. Quantity per application Number of initial depot serviceables 12. Number of initial depot unserviceables 13. Number of initial war reserve (retail) serviceables 14. Number of initial war reserve (retail) 15. unserviceables 16. Total parts in retail ASL/PLLs on Day-1 Distribution schedule of parts deployed after Day-1 17. (by 5-day interval)

A-2. CHANGES IN PARTS DATA BASE INPUT. Extended PARCOM shares elements (1) through (11) of Table A-1 with basic PARCOM. However, while basic PARCOM at Day 1 emplaces in the theater all the available for each part, stock, extended PARCOM allows for distribution of that stock over time, through the inclusion of the additional data elements (12) through (17).

A-3. SCENARIO DATA BASE INPUT. In addition to the parts data base, extended PARCOM inputs the scenario data listed in Table A-2.

A-4. CHANGES IN SCENARIO DATA BASE INPUT. Relative to the parts data base used in basic PARCOM, the extended PARCOM includes essentially all basic PARCOM scenario input, but adds the following data/capabilities:

a. Depot distribution and lag times. All basic PARCOM parts were front loaded.

b. Partial-substitution policy specification.

c. Options to specify the list order of part requirements. Basic PARCOM listed requirements only in order of decreasing part unit cost.

d. Options to do only a capability assessment of current inventory under several different partial-substitution policies. Basic PARCOM did a current inventory capability assessment only for a no-substitution policy.

Table A-2. Data Elements for the Scenario Data Base

Scenario Specification Data

- Case identifier
- Length of war
 Flying program
- Aircraft deployment schedule
- Aircraft losses

Scenario Constraint Data

- Cost limit (for constrained cost)
- Aircraft availability constraints (minimum daily availability)
- Maximum flying hours per aircraft per day

Additional Parts Data

- Order ship time offset
- Maximum essentiality code for part to be processed
- Lag time before initial depot serviceables are sent to retail
- Duration of time required to distribute initial depot serviceables to retail

Part Replacement Policy Specification Data for Requirement Calculations

- (1st Option) Number of parts in full-sub parts set and the part numbers of the parts designated as full-sub
- (2nd Option) Screening limits on depot cycle time, NRTS rate, retail repair time, and failure rate. A part type with parameters exceeding any screening limit is selected for the full-sub set.

Part Replacement Policy Specification for Current Inventory Capability Assessment

Number of parts in each full-sub parts set and the part numbers of parts . designated full sub

Print/Calculate Options

- Options to print various input/output lists
- Options to omit requirements calculations and only do capability assessment of current inventory
- Option to select the order in which part requirements are listed in output--either by decreasing part unit cost or by decreasing amount of requirement

Tuning Parameters

- Desired closeness of "flying hours flown" convergence during capability assessment
- Maximum number of iterations used to calculate "flying hours flown" during capability assessment
- Increment step size used during partial-substitution requirements calculations

Miscellaneous

Designation of (up to 100) part types for which cumulative requirements through each scenario day will be listed

APPENDIX B

EXTENDED PARCOM - PROGRAM SOURCE CODE

MAIN PROGRAM	pages	B-3 thru B-11
SUBROUTINE CCCAP	pages	B-13 thru B-14
SUBROUTINE CCLIST	page	B-15
SUBROUTINE NCRNC	page	B-17
SUBROUTINE UCCAP	pages	B-19 thru B-20
SUBROUTINE UCRQRS	pages	B-21 thru B-23
FUNCTION MAXC	page	B-25
FUNCTION SR	page	B-27
SUBROUTINE DIST	page	8-29

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(NOT USED)

MAIN PROGRAM

1234

56789

1111111111222345678901234567890

414345

46748

60

71

80

NAME: PARCOM-X TYPE: MAIN PROGRAM CCCC WRITTEN BY: WALTEP BAUKAN/AUTOVON -295-1662 AT: US ARMY CAA/612D VOODMONT AVE,BETHESDA,MD 20814 CCCC PURPOSE: THE PAPCON-X (PARTS REQUIREMENTS AND COST MODEL-EXTENDED) IS USED TO GENERATE COST EFFECTIVE MIXES OF SPARE PARTS REQUIRED TO ACHIEVE A FLYING PROGRAM/AVAILARILITY OFJECTIVE UNDER A USER-SPECIFIED -PART REPLACEMENT POLICY (EITHER FULL, PARTIAL OR NO SUBSTITUTION) -(PURCHASE) COST CONSTRAINT 000000 IN ADDITION, THE PROGRAM ALLOWS THE CAPABILITY ASSESSMENT OF AN AIRCRAFT FLEET BASED ON A USER-SPECIFIED SPARES INVENTORY APPLIED UNDER A VARIETY OF USER-SPECIFIED PARTS REPLACEMENT POLICIES CCC ARGUMENTS: NOT APPLICABLE CALLED BY: NOT APPLICABLE CALLS -SUBROUTINE MAXC: ORDERS PART TYPES IN DECREASING OFDEP OF UNIT COST -SUBROUTINE CCCAP: PERFORMS A FLEET CAPABILITY ASSESSMENT BASED ON A SPARES STOCK EQUAL TO THE CONSTRAINED COST SOLUTION AND/OR CURRENT INVENTORY -SUBROUTINE CLIST: PINTS SELECTED CONSTRAINED COST SOLUTIONS -SUBROUTINE DIST: DISTRIBUTES PARTS TO THEATER OVER 5-DAY INTERVALS -SUBROUTINE UCRCPS: COMPUTES A COST-EFFECTIVE REQUIPEMENTS MIX BASED ON THE UNCONSTRAINED COST SOLUTION BEING STOCKED -SUBROUTINE UCCAP: COMPUTES FLEET CAFABILITY ASSESSMENT BASED ON THE UNCONSTRAINED COST SOLUTION BEING STOCKED FILES USED : INPUT - UNIT 1D (PARTS DATA) - UNIT 11 (SCENARIO DATA) OUTPUT - UNIT 6 (PRINT) DIMENSION ALR(120), DC(3D0), 10AY(61), PT(24), AM(61), DSER(30D), NAC(61), WRES(300), DAY10(300), FR(300), OST(300), XRNCS(300), BC(100); DUNSER(300), NFH(61), WRESU(300), + + ÷ ZNPT(300) ZLOŠS(61), COMMON AC(12D), ALLOWB(12D), CDMCA(30D), CDMCA(30D), DCOSTF(12D), DCOSTF(12D), FHR(12D), FHR(12D), IFS(30D), IFS(30D), NP, • ZLOSS(61), ACL . AHSN(300). BCY(300). ADESC (300), ASURV (120), BF (300), ALLOW1(120), AVAV6(6), +++ AVAV6(6), CASE, CMINT, DCOST1(300), DMP(300), FHAPD(3,120), IFHC(120), ٠ BCY(3DD), CF(3DD), COST(3CD), DCY(7DD), FHA(12D), ICOST, IMSEL, IRC(3DC), NP1, SRMAY1(3DD), TSTK(3DD), BF (300), CL, CRNCS(300), DF (300), FHM, IOCC(2), INS(300), IRS(300), NP2, RNC(300), STK(300), TSUMB +++++ ٠ ++++ INT. ISHORT, + SUMB(120), ADSC, AMSN, CASE . 100 200 INS(1)=0 2Z=0. KNTC=1 300 READ (11,9000) ADDOST,CONVF,IESS,DLAG,DDIS NP=0 NP1=0 READ (11,9100) NFS IF (NFS.LT.0) READ (11,9200) ZDCY,ZNRTL,BREPL,FRLIM IF (NFS.LE.0) 60 TO 400

B-3

82 83 84 85	400	READ (11,9100) (IFS(J),J=1,NFS1 WRITE (6,9100) (IFS(J),J=1,NFS) READ (11,9300) CASE READ (10,9300) CASE
86 87 88	500	T=0 T=0 READ (10,9500,END=1303) 21,22,23,24,25,76,27,28,29,IES,INIT READ (10,9600,END=1303) DSRV,DUNS,WRS,WRU,DAY1 PFAD (10,9700,END=1302) DSRV,DUNS,WRS,WRU,DAY1 PFAD (10,9700,END=1302) DSRV,DUNS,WRS,WRU,DAY1
90 91 92 93		READ (10,990,END=130) ANSC READ (10,990,END=130) (PT(K),K=1,24) READ (10,940,END=130) IF (IFS,GT,IFSS) 60 TO 700
94 95 96 97		2T=Z3*ACDOST ZXD=2.*ZT*Z7 Z2C=Z2/100. Z4F=Z4/1000000.
98 99 100 101		Z5N=Z5/100• Z10C=ICPA Z8B=Z8/100• Z9D=Z9/100•
102 · 103 104 105		IF (MUDINP+1,SUI.NE.U) GO TO 600 WRITE (6,10000) CASE WRITE (6,10100) WRITE (6,10200)
107 108 109	600 700	IF (24-5E0000001) GC TO 800 WRITE (£,10400) 21,405C,2°C,23,24F,75N,26,2X0,27,28P,790,2100,1ES 1=1 50 TO 500
111 112 113 114	800	NP=NP+1 STK(NP)=WRS+DAY1 BCY(NP)=Z6 DCY(NP)=D+
115 116 117 118		IF (Z5N.GT.D.) DCY(NP)=ZXP ZNRT(NP)=Z5N CLASS(NF)= NO SUB* IF (NFS.GE.D) 50 T0 900
120 121 122	900	CLASS(NF)="FULL SUB" IF (BCY(NP).LE.BREPL.AND.DCY(NP).LE.ZDCY.AND.ZAF.LE.FRLIM.AND.ZNRT (NP).LE.ZNRTL) CLASS(NP)="NO SUB" GO TO 11DD TE (NES ED D) ED TO 1100
124 125 126 127	920	IF (TFS(L).KE.NP) GO TO 1000 CLASS(NP)= FULL SUB GO TO 1100
128 129 130 131	1000	CONTINUE WRITE (6,1050C) NP, Z1, ADSC, Z2C, Z3, Z4F, Z5N, Z6, ZX0, Z7, Z88, Z90, Z100, I +ES, CLASSINPJ, STK(NP) OSI(NP)=ZI
132 133 134 135		A M SN (NP 1=21 COST (NP)=22C FR (NP)=24F BC (NP)=28B
137 138 139		DPA(NP)=200 ADESC(NP)=ADSC DSEP(NP)=DSRV DUNSEP(NP)=DUNS
141 142 143 144		WRESUNPJEWRS WRESU(NPJEWRU DAYID(NPJEDAYI D0 1200 L=1
145 146 147 148	1200	PIDEPINP,LJ=PI(L) IF (NFS.GE.D.OR.CLASS(NP1.EQ." NO SUB") GO TO 500 NP1=NP1+1 IFS(NP11=NP EO TO 500
150 151 152 153	1300	ĬĬ=ŃP+Ĭ IF (NFS+GE+D) NPI=NFS Write (6,10600) II,NP Read (11,10700) CLNCT,CLNCT,LIMIT
154 155 156 157		READ (11,10800) FHH,NW,ISEL,IORD,IOPT1,IOPT2,IOPT3,IOPT4,IOPT5,IPR +T,IPRT1,INT IF (NP1.60.0.0R.IPRT1.LF.M) GO TO 1500 D0_1400_111,NP1
158 159 160 161 162		II-IF311 IF (MOD(I-1,50).NE.D) 60 TO 1400 WRITE (6,10000) CASE WRITE (6,10900) KNTC WRITE (6,10200) WRITE (6,10200)

```
14D0 WRITE (6,11DDD) II, AMSN(IT), ADESC(II), COST(II), FR(1)), ZNRT(II), BCY

+111, CCY(II), PC(II), CC(II), STK(II)

1500 NF2=0

D 1800 K=1, NP

If (Fri.C.D) 60 TO 1700

0 16CD CONTINUE

1700 NP2:NP2+1

1800 CONTINUE

1700 INS(NP2)=K

1800 CONTINUE

II=INS(I)

II=INS(I)

II=INS(I)

II=INS(I)

II=INS(I)

VRITE (6,1000D) CASS

WRITE (6,1000D) CASS

WRITE (6,1000D) CASS

WRITE (6,1000D) CASS

WRITE (6,1000D) II, AMSN(II), ADESC(II), COST(II), FR(1I), ZNRT(II), BCY

+(II), OCY(II, BC (II), CC(II), STK(II)

2000 RETE (6,1000D) II, AMSN(II), ADESC(II), COST(II), FR(1I), ZNRT(II), BCY

+(II), OCY(II, BC (II), CC(II), STK(II)

2000 RETE (6,1000D) INACEP

READ (II, 9100) NACEP

READ (II, 9100) NFHOAY

READ (II
                                                                                                                      FHA(J)=NFH(I)
FHR(J)=NFH(I)
CONTINUE
READ (11,91CD) NLCAY
READ (11,91CD) (ICAY(I),I=1,NLDAY)
READ (11,130CD) (ZLOSS(I),I=1,NLDAY)
DD 260C I=1,NLDAY
K1=ICAY(I)
K2=ICAY(I+1)-1
IF (I.EQ.NLOAY) K2=NW
OD 2500 J=K1,K2
ALR(J)=ZLOSS(I)
CONTINUE
READ (11,91CO) NMDAY
READ (11,91CO) (ICAY(I),I=1,NMOAY)
READ (11,91CO) (ICAY(I),I=1,NMOAY)
DD 26CC I=1,NMOAY
K1=ICAY(I)
I CONTINUE
READ (11,91CO) (AM(I),I=1,NMOAY)
DD 26CC I=1,NMOAY
K1=ICAY(I)
I CONTINUE
REAC (11,91CO) INSEL
REAC (11,91CD) INSEL
REAC (11,91CD) INSEL
REAC (11,91CD) (ICAY(I),K=1,IMSEL)
IF (IFRTI+LE.0) 60 TO 33DD
ZCOST=0.
D0 3000 K=1 aNP
                         250D
2600
                      2700
                         2800
            KLEWU 1117100, 117110, 11710, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1171, 1
```

164

1 8D

188

199012345678990012345678901012345678990012345678990012345678900123456789001214567890010123456789001012

22234567890123456789E

746		TE 4H00((K-1)+3,60), NE-0) 60 TO 3100
247		WRITE (6,10000) CAST
248		
249		WRITE (6.11900)
250		WRITE (6,12000)
251	-	WRITE (6,10300)
252	3100	WRITE (6,12100) K,AMSN(K),ADESC(K)
253	3200	WRITE (6,12200) (PTDEP(K,L),L=1,24)
254	3300	DU 3500 KEI NP
222		
250		
558		CALL DIST (IFDAY, TEDAY, PANT, K)
259		
260		DREP=DCY(K)-2.+OST(K)
261		IF (D°EP+LT+1+) DREP=3+000
262		ILDAY=OST (K)+OREP
263		
264		CALL DIST (IFDAY, ILMAY, DANT, K)
265		A MITE (] - T A NHI (K)) + WH C SU (K) + (] - BC (K))
200		IF UAT-I TP (DOUM) IT 1 . APU/MI-1
260		
269		
270		CALL DIST (IFDAY, ILDAY, CANT.K)
271		AMT=ZNRT(K)*WRESU(K)*(IDC(K))
272		IFDAY=1.+2.+DST(K)
273		ILDAY=2++0ST(K)+0REP
274		DAMTSANT/DREP
275	•	CALL PIST (IFDAY, ILPAY, PANT, K)
276		IF (IFRTI-LE-D) GO TO SOD
270		17 TTE 4 10703 (455
279		
280		
281		WRITE (6,12000)
282		WRITE (6,10300)
283	3400	IF (DCY(K)-GE-+000I+OR+PAMT+LE++001) WRITE (6,12100) K,AMSN(K),A
284	•	• DESC(K)
Z85		IF (DCY(K)+LT++DDD1+AND+DAMT+GT+DD1) WRITE (6+124CD) K+AMSN(K)+A
280		UESCINI Hotel (1990) (proprint 1) 171 200
288	1500	CONTINUE (0,12200) (PIDLP(F,1),1,24)
289	2200	
290		00 37nn J=1 NV
291		IF (MOD(J-I-S1)-NE-D) 60 TO 3600
292		WPITE (6,10000) CASE
293		WRITE (6,12500)
294		WRITE (6,12600) ADDUST, CONVE, LIMIT, JESS
322		WRITE (6,12700) FRM, CLNCR, CLNCT
290		WRITE (6,12800) 20031
208		
299	3600	
300	3700	WRITE (6.13100) J.AC(J).FWR(J).AVM(J).ALP(J).CALR
301	3800	WRITE (6,1000C) CASE
302		WRITE (6,13200)
303		IE (ISEL-EC-D) WRITE (6,17300) ISEL
304		IF (ISEL-ED-1) WRITE (6,13400) ISEL
202		IF (ISELLEQ 2) FRITE (6,13500) ISEL
300		IF INFS OLD UT WAITE TO ISOUUS AFSILLT CARACLEDAELE FREIN
308		TE ATORDULE OF WRITE 16.138003 TORD
309		IF (IDEC.GT.O) WRITE (6.17900) TORD
310		IF (IOPTI-LE-C) WPITE (6.14000) IOPT1
311		IF (IOPTI-6T-C) REITE (6,14100) JOPT1
312		IF (IDPT2-LE-D) WPITE (6,14200) IOPT2
213		IF (10F12.6T.U) WPITE (6,14300) 10P12
214		IF (10013-12-0) WRITE (6,14400) 10013
212		TE TIGETALF.OS WATTE TE 14007 TOPIS
317		1F 110FT4.5T.01 URT TE 16.147001 10PT4
318		IF (IOPTS-LE-O) WRITE (6.14800) IOPTS
319		IF (IOPTS.GT.D) WPITE (6,14900) IOPTS
320		IF (IPPT-LE-D) WRITE (6,15000) IPRT
321		IF (IPPI.GT.D) WRITE (6, ISIOD) IPRT
322		IF (IFRI].LE.D) WRITE (6,ISZOD) IPRTI
323		IF TIMELADIANI THI TOTION INTICONSTRUCT
325		RAIC 109134007 INT9181
326	3900	bčošíří i j=0.
327		DO 4000 1=1,NP

B-6

400D DOD(I)=CDST(I) DOD(I)=CDST(I) KNT=0 NDUMMY=NP DO 4300 K=1,NP CALL MAXC (NDUMMY,NDUT1 IRC(K)=NOUT II=IRC(K1 IF (NF1.LE.01 GO TO 4200 DO 4100 L=1,NP1 IF (IFS(L).EQ.II1 GO TO 4300 CONTINUE KNT=KPT+1 INS(KTT)=II DOD(II)=-1. . 4200 4100 CONTINUE 4200 KNT=KTT=1I 4300 D004IJ=-1* IF diPrI*L=C=C1 60 T0 4600 D0 4500 K=1,NP D0 4800 I=1;24 4400 SUM=SUM+PTOEP(II]+11 SUMT=SUM+OSER(II)+CUNSEP(II)*(1.=0C(II))*VRES(II)*0AY10(II)*(1.= * ZNRT(II))*VRESUIJ*CI) 60 T0 4500 WRITE (6,ID00C1 CASE . BF (J)=(1.-BC(J))*(2*RT(J))*CF(J) OF (J)=(1.-DC(J))*(2*RT(J))*CF(J) IF (IOPTI.LE.OI GO TO 7600 INO1=1 INO2=2 IF (ISEL.EQ.2) GO TO *0C0 INO1=1+ISEL INO2=1+ISEL S000 DO 7500 INO=IND1.IND2 ACL=0. CL=CLNCT ICOST=0 IF (IND.E0.2) CL=CLNCR CALL UCCAP (INO) IF (CL.LE.0.1 GO TO 7500 IW=NW UCNS =0. FRAC1=0. IF (NF2.EC.0) GO TO 660" DD 5100 J=1.NP2 II=INS(J) S100 UCNS=UCNS*COST(II)*PNCS(II) WRITE (6.15700) DD 52C0 J=1.NP S200 TRNCS(J1=RNCS(J1 CL1=CL CNC=CMINT-CL IF (IND.E0.1) WRITE (E.158001) IF (IND.E0.2) WRITE (E.158001) IF (IND.E0.21 WRITE (E.159001) GO TO 7500 S300 IF (CNC.GT.CCOST1(NW)) GO TO 5400 CL1=UCNS CL1=UCNS

369 370

410	CL2=CL-CL1
412 5400	IF (NP1+EQ+0) 60 TO 5600
413	00 5500 J=1,NP1
415 5500	TRNCS(II)=D.
416 5600	CL2=0.
918	CNC=UCNS-CL1
419	II=INS(J)
421	C=TFNCS(II)+CCST(II)
423	TRNCS(II)=TRNCS(II)-CNC/COST(II)
929	IV=NV 60 T0 6200
426 5700	TRNCS(II)=0.
427	CNC-C Continue
429	60 TO 6100
431	00 6000 I=1,NW
4 32 8 3 3	IF (DCOST1(I).6T.CL2) GO TO 6000 IFCC=I
4 34	BCL=DCOSTINI
435 6000	WRITE (6.16000)
437 6100	IF (CL2.GF. DCOST1 (NW)) WRITE (6,16100)
4 30	IW=NW
441	NWIFCC CALL PCROPS (IND.IOPTE.TOPIS.TORD)
442 6200	WRITE (6,16300) CL,CL1
444	NW=IW
445	00 6300 I=1,NP2
447 6300	RNCS(II)=TRNCS(II)
448 6400	UU 6400 I=1.NF TRNČS (IJ=RNČS (I)
450	TOT=0.
452	TOT=TOT+COST(I) *RNCS(7)
453 6500	IP=0
4 55	CALL CCCAP (IND,LIHIT,CONVETTEH,KNTC, IP, FNC)
457	WRITE (6,16500) FRACI
459	NW=ICCC(IND)
460 461 6700	00 67C0 I=1.NV FHALIJ=FHRLIJ
462	CALL UCROPS (IND, IDPTA, IDPT5, IDRD)
464	XRNCS(J)=RNCS(J)
465	NV IV TP = C
67 6800	RNCS(J)=RNCS(J)+STK(J)+(IND-1)
469	FRAC2=FNC
470	WRITE (6,16600) FRAC2 TE (FPAC1,15,FRAC2) 60 TO 7100
472	D0 6900 J=1.NP
474	IG=1
475	ACLETOT CALL CELIST (IG.IORD.IND)
477	
479	IP=1
480 481	CALL CCCAP (IND,LIMIT,CONVF,TTFH,KNTC,IP,FNC) 60 TO 7400
482 7100	
484 7200	
485	CALL CCLIST (I6,IOR",IND) 00 7300 J=1.NP
487 7300	RNCS(J)=XRNCS(J)+STK(J)+(IND-1)
189	CALL CCCAP (IND, LIHIT, CONVE, TTEH, KNTC, IP, FNC)
491 7500	ICOSTEC

•

	7600	00 7700 K=1 NP FNCS (K)=STK (K) IP=1
		INDE2 CALL CCCAP (IND,LIMIT,CONVE,TTEH,KNTC,IF,ENC)
	7800	00 7900 K=1,NP RNCS(K)=STK(K)
	7960	IFS(K)=C READ (11,9100,END=16730) NPIFS NPINS
		IF (INFIESTED) 60 TO 8200 IF (NPTFS-LE-0) 60 TO 8200 NP1=NPTFS
		READ (11,9100) (IFS(I),I=1,NPTFS1 NP2=0 D0 8100 k=1.NP
	8000	DO 8000 I = 1, NP1 IF (IFS(I).EQ.K) GO TO 8100
	0000	
	8200	GO TO 8500 NP2=NPTf:S
		READ (11,9100) (INS(I),I=1,NPTNS) NPI=0
		DO 6400 K=1 NP2 DO 6300 I=1 NP2
	8300	CONTINUE
	8400	IFS(NPI)=K
	8500	IF (10PT2+LE+C) GO TO 8700
		II=1FS(I) IF (PC0(I-1,50).NE.0) 60 TO 8600
		WRITE (6,1000C) CASE WRITE (6,10900) KNTC
		WRITE (6,10200) WRITE (6,10300)
	8600	WRITE (6,11000) II,AMSN(IT),ADESC(II),COST(II),FR(II),ZNRT(II),BCY +(11),DCY(I1),BC(II),DC(II),STK(II)
	6700	1F (10PT3-LE-D) 60 TO 8900 D0 8800 I=1 NP2
		II=INS(I) IF (KOD(I-1,5C).NE.D) 60 TO 8800
		WRITE (6,1110C) KNTC
÷		WRITE (6,10200) WRITE (6,10300) WOITE (6,10300)
	6000	+(II),DCY(II),BC(II),DC(II),STK(II)
	8900	CALL CCCAP (IND LIMIT, CONVE, TTEH, KNTC, IP, ENC) KNTC=KNTC+1
	9000	60 TO 7000 FORMAT (2F5+2+15+2F5+0)
	9100	FORMAT (1615) FORMAT (F5.0.F5.3.F5.0.F10.6)
	9300 9400	FORMAT (1X, A16) FORMAT (//)
	9500 9600	FORMAT (2X,A15,F9.0,5X,F3.0,F5.0,5F3.0,11,10X,I51 FORMAT (/,5F6.0,/)
	9700 9800	FORMAT (12) Format (A16)
	9920	FORMAT (10F10.0) FORMAT (1H1.30%, *CASE= *, #16)
	10100	FORMAT (//, " ITEMS RANK OFDERED IN NORMAL INPUT ORDER") FORMAT (/, " PART", 5X, "MSN", 14X, "DESCRIPTION", 7X, " COST OST FAIL","
	10300	+ RT NRTS BCY DCY DRT BCON DCON QPA ESS CLASS INIT STR*) FORMAT (/)
	10500	- UKHAI - 17A BAIDACABAIDALE + UBLE + UBLE + OBLE + CABLE - 2 - 12 - 2 - 14 - 14
	10500	+0,15,1X,48,510,13 FORMAT (* 107AL NR PARTS=*.I%.* NR HSED=*.I%1
	10700	FORMAT (1X, F14.0, F15.0, T5) FORMAT (1X, F9.1, 15, 5X, 1C15)
	10900	FORMAT (7/, * FULL SUB ITEMS FOR POLICY*,13) FORMAT (1X,14,4X,A16,2X,A16,F8.0,3X,F8.6,F5.2,2F5.0.5X,2F5.2.1X.10

574	+X, *FULL SUB *, F10.0)	
575	11100 FORMAT (//+: NO SUB ITEMS FOR POLICY + 13)	
577	11200 FORMAL 112, 14, 14, 14, 14, 14, 14, 14, 10, 22, 4, 10, FO, 0, 9, 3, 4, FO, 0, 0, 7, 3, 4, 2, 5, 4, 2, 5, 4, 12, 12, 12, 14, 14, 14, 14, 14, 14, 14, 14, 14, 14	10
578	11300 FÖRMAT (16F5-1)	
579	11400 FORMAT (16F5-2) 11500 FORMAT (4.1047.) OFPLOYEO.)	
581	11600 FORMAT (* RANK PAPT ", "X, "MSN", 12X, "DESCRIPTION", 13X, "COST", "	CL
582	+ASS',2X, DEERV DUNSE WEERV WEUNS DAY1 DAY2- TOT NC')	
585	11/00 FORMAT (213-53, A10, 53, A10, 13, 14, 01, 14, 00, 13, A0, 13, 57, 60, 0, 27, 0, 0, 1 11800 FORMAT (21, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0,	
585	11900 FORMAT 1/, -5 -10 -15 -20 -25 -30 -35 -40 -4	5 *
586	•, • -50 -55 -60 -65 -70 -75 -80 -85 -90 -95*,	,
588	12000 FORMAT (* -105 -110 -115 -120*)	
589	12100 FORHAT (15+2X+A16+2X+A16)	
591	12300 FORMAT (14,2000-00) FOR (FOR DEPOT STKS) PARTS DEPLOYED BY INTERVAL	
592	+)	-
593	12400 FORMAT (15,7X,A16,2X,E16,X,** WARNING+ DEPOT UNSERV STK W/ DEP*	•
595	12500 FORMAT (//, 10X, SCENARIO INPUT DATA SUMMARY)	
596	12600 FORMAT 1//, 5X, CST CFFSET= ", F6.1," DAYS DESIDED CONVERGECE="	5
598	1270 FORMAT (4.5%, " MAX FLY HES/ACFT/DAY=",F5.1.4%, "ADD-ON COST LT".	M.
599	+11=*, F12.0, 3%, *TOTAL (INIT INV=0) ROMT (OST LIMIT=*, FI3.0)	
600	12800 FORMAT (7,5%, COST OF CURRENT INVENTORY=",F14.0) 12900 FORMAT (7,5%, CUM ACFT DOGRAM MIN PRO ACFT CUM ACFT")	
602	13000 FORMAT (7X, OLY OEPLOYED FLY HRS AVAIL LOST ,7X, LOST	• >
603	13100 FORMAT (5X, 15, F11.0, F10.0, F10.2, F8.1, F)1.1)	
605	13200 FORMAT 1/////SX.*ISEL*.13.* ++ ONLY THE TOTAL (INT STK=0)*.* RI	H
606	+TS ARE COMPUTED IN THIS RUN ++*)	
607	I3400 FORMAT (////,5X, "ISEL=",I"," ## ONLY THE RESIDUAL(INIT STK=CURR +* TNY) FORMTS APP COMPUTED IN THIS PUN ##*)	
609	ISSOO FORHAT (1/1/151 ISELE JIJ BOTH THE TOTAL (INIT STK=0) ANO.	
610	+RESIDUAL(INIT SIK=CURV INV) REMIS ARE COMPUTED IN THIS RUN ++*)	
612	+TO A DEPOT REPAIR CYCLE EXCEEDING .F12.C. DAYS OP NRTS . EXCEE	IC
613	•NG* F6-3 //15X, OP RETAIL REPAIR TIME EXCEEDING*, F8.0, OR FAI	LU
615	13700 FOFMAT (//.5X.*NFSZ*13.* ++ FULL SUB SET IS SPECIFIED BY INPUT	•)
616	13000 FORMAT 1/2 5% 10R0 = 13, * ** COMPUTED ROMTS LISTS WILL BE IN *	• •
618	+ORDER OF DECREASING UNIT COST OF PARTY 13900 FORMAT 1/2-51 TORDEY 13-13- ** COMPUTED ROMTS LISTS WILL BE IN *	
619	+ORDER OF DECREASING REMT AMOUNT FOR PART	
620	140CD FORMAT (//,5%,"IOP11=",13," ** ONLY ASSESSMENT CASES WILL BE ',	0
622	14100 FORMAT (//,5%,"IOPT1=",13," ** BOTH ASSESSMENT AND ROMT CASES",	•
623	+VILL BE DONE IN THIS RUN")	
625	+ASSESSMENT CASES WILL NOT BE PRINTED 1	
626	14300 FORMAT 1//,5X, "IOPT2=", I3, " ++ THE FULL SUB PART SETS USED IN",	•
628	14400 FORMAT 1/1.5X." IOPIZE JULE BE PRINTED THE NO SUR PART SETS USED IN"."	2 4
629	+SESSMENT CASES WILL NOT BE PRINTED)	
631	14300 FORMAN 177,524,10713-1,134, WE IME NO SUB PART SETS USED IN."	13
632	14600 FORMAT 4/2, 541 TOPTA - 131 THE UNCONSTP COST RONTS LISTS	¥.
633	+ILL NOT BE PRINTED(401 #RF COMPUTED)*) TATOR FORMAT (4/55, *TOPIGE', 13, * ** THE INCONSTP COST PONTS (1555, *	W.
635	+ILL BE PRINTED ')	
636	14800 FORMAT (//,5%,"IOPTS=",J3," +* THE CUM ROMT BY DAY COST LISTS",	•
638	14900 FORMAT (//, 5%, "10" 5=", 13," ++ THE CUN RONT BY DAY COST LISTS",	
639	+VILL BE PRINTED 1	
641	+VILL NOT BE PRINTED *)	
642	15100 FORMAT 1/2.5X TPRT="13," ++ THE SCENARIO INPUT DATA SUMMARY",	•
643	+VILL BE PRINTED *) 15200 FORMAT (4/.5%, "TORTIS", TT," AR THE FILL SUB AND NO SUR PART *."	s
645	+ETS (FOR ROMT CASES) WILL NOT BE PRINTED . / ISX, NOR WILL ", "THE	Ĩ
646	+NPUT-CRCERED AND COST-OPDERED PARTS INPUT LISTS ")	
648	+ETS (FOF ROHT CASES) VILL BE PRINTED +/+13X + AS WILL + THE INPU	1-1
649	+ORDERED AND COST-ORDERED PARTS INPUT LISTS *)	
651	ALL TEST AT INTERVALS OF J3. (ALLOWARE NMCS ACFT) +	H I
652	ISSOD FORMAT (/, ' ITEMS RANK ORDERED BY DECREASING PART COST')	
653	15600 FORMAT (1H1) 15700 FORMAT (77, TARA CONSTRAINED COST SOLUTION CVALUATION	σт
655	+ *** * * * * * * * * * * * * * * * * *	

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15800 FORMAT (1H1.//.10x, THE UPCONSTP COST TOTAL RCMT SOLUTION IS', AL +S0 THE CONSTF CCST TOTAL FOMT SOLUTION') 15900 FORMAT (1H1.//.10x, THE UPCONSTP COST RESIDUAL POMT SOLUTION IS', ALSO THE CONSTP COST RESTDUAL POMT SOLUTION') 16000 FORMAT (//.10x, ALL INO SUB) PARTS ARE AFFORDABLE IN CONSTRAINED *, COST SOLUTION I') 16100 FORMAT (//.10x, ALL FULL SUB PARTS APE AFFORDABLE IN CONSTRAINED *, COST SOLUTION I') 16200 FORMAT (//.5x,F12.0,3X, AFPROXIMATED BY',F12.0, CUM FULL SUB', *PART CCST THRU DAY',I&, IS USED TO BUY FULL SUB PARTS') 16300 FORMAT (//.10x, CONSTP COST LIMIT=',F12.0,3X, CF WHICH',F12.0,3X, *CAN BE USED TO BUY IN SUB PARTS ARE AFFORDABLE IN CONSTRAINED *, COST SOLUTION I') 16400 FORMAT (//.10x, NO FULL SUB PARTS OF THE UNCONSTR COST SOL') 16400 FORMAT (//.10x, NO FULL SUB PARTS ARE AFFORDABLE IN CONSTRAINED *, COST SOLUTION I') 16500 FORMAT (//.10x, NO FULL SUB PARTS ARE AFFORDABLE IN CONSTRAINED *, COST SOLUTION I') 16400 FORMAT (//.10x, NO FULL SUB PARTS ARE AFFORDABLE IN CONSTRAINED *, COST SOLUTION I') 16400 FORMAT (//.5x, THE FIRST CONSTR COST SOL YIELDS AN AVG FRAC', PGM *FLY HRS ACH=',FS.3) 16500 FORMAT (//.5x, THE 2ND(SUSTNBLTY) CONSTP COST SOL YIELDS AN *, AVG * FRAC FH ACH=',FS.3) 16700 END 665589C12234566789011234 6666666666666666777234

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(NOT USED)

SUBROUTINE CCCAP

SUBROUTINE CCCAP (IND.LIMIT.CONVF.TTFH.KNTC.JP.FNC) CCCAP TYPE: SUBROUTINE CCC NAME: CCCAP PURPOSE: THE CCCAP (CONSTRAINED COST CAPABILITY ASSESSMENT) SUBROUTINE COMPUTES FLEET CAPABILITY ASSESSMENT (AVG AVAILABILITY, FRACTION FLYING PROGRAM ACHIEVED, FGM FLYING HSS /ACFT/DAY) BASED ON THE CONSTRAINED COST SOLUTION BEING STOCKED IN THE WAR RESERVE PURPOSE: THE CCCAP 0000 CALLED BY: MAIN PROGRAM CALLS -FUNCTION SR: COMPUTES CUMULATIVE NET DEHAND THRU A SPECIFIED DAY FOR A SPECIFIED PART FILES USED : INPUT - NONE DUTPUT - UNIT 6 (PRINT) 10N AC(120), ALLOWB(120), AVH(120), COMDA(300), CNCS(300), DCOSTF(120), DOD(300), FHP(120), IFS(300), IFT(100), NP. COMMON ALLOWI(120), AVAVG(6), CASE, CMINT, CCCST1(300), CMD(300), FHPAPD(3,120), IFHC(120), INT, ISHORT, NH. ACL, AMSN(300), BCY(300), CF(300), COST(300), DCY(300), ADESC (300). ASURV (120), BF (300), ++ +++ CL + CRNCS(300), DF(300), 4 . DF(300), FHM, IDCC(2), INS(3⁻0), IRO(300), NP2 RNC(120), STK(300), TSUMB FH1(120), ICOST, IMSFL, IRC(300), + +++ + NP1 CP#[3001, SRMAX1(300), TSTK(300), NP, PTČEP(300,24), SM(120,100), TRNCS(300), ++ NU . RNCS (300) . SUPB (120) . + SH (120,100), SR*
 TRNCS(300), TST
 DIMENSION
 OMOT(300), FHN
 CHARACTER*16
 ADESC, ADS
 BMAX:00, AVVG(1):00, TFHNC:00, TSURV:00, TFHNC:00, TSURV:00, TSURV:00, TSURV:01)
 DO 100 I=1,NW
 TSURV:00, TSURV:ASURV(I)
 SUMB(I):00, SUB(I):00, SUB(I): • FHNC(120), FHNZ(120) ADSC. AHSN . CASE XX=ASUPV(1) TAV=0. OD I200 I=1,NW IA=(I-1)/5+1 DD 3CD J=1,NP RNCS(J)=RNCS(J)+(IND-1)+PTDEP(J,IA)/5. IF (I.GT.1) XX=RNC(I-1)+ASURV(I-1)+AC(I)-AC(I-I) FHA(I)=AMIN1(XX+FHM,FHR(I)) INOX=0 IF (NP2.E0.D) 60 TD 60D DD SCC K=1,NP2 II=INS(K) XX=0HOT(II) DMDT(II)=SR(I,NIT,YX) 300 400 XX=DHOT(II) DHOT(II)=SR(I,II,YX) ZP=CHOT(II)=RNCS(II) SUMP(I)=SUMB(I)+AMAX1(0.,ZP) IF (NP1.EQ.D) GO TO 80D BMAX=0. 00 70C K=1,NP1 II=IFS(K) XX=CHOT(II) DVCT(II)=SR(I,II,YX) BOFCS=(CHOT(II)=RNCS(II))/QPA(II) IF (BDFCS.LE.0.) %OFCS=0. BMAX=AMAX1(BMAX,BDFCS) CONTINUE AUNCS=AMAX1(0.,ASURV(I)=SUMB(I)=BMAX) FHNC(I)=AMIN1(FHR(I),AUNCS+FHM) FHPAPC(3,I)=AMIN1(FHM,FHR(I)/(AUNCS+.OI)) FHNZ(I)=FHNC(I)/(FHP(I)+.000001) 500 600 700

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B-13

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1 37	240	Ő,	FO	RH.	AT	(6)		•0	AY	•	7	ι,	• 5	Ŭ	e •	• -	3 X	• •	R	ĒG	1	V	AI	Ĺ	٠,	6 X	, "	0	AY	٠,	67		• 5	UB	••	11	×, '	SU
139	250	0	FO	RH	AT	(5>	(I	14	.5	X	F	5.	3,	7	× ,	F S	5.	3,	5	X,	1	4 ,	4)	¢.,	FS	. 3	16	x	2F	8.	1		_					
140	260	0	FOEN	R M D	AT	(1	•	A	VG	1	V.	AI	L'		EX	,1	5	• 3	5 9	7 X		5	•	5.	12	х,	FS	•	5,	10	X	F	5.	1)				

.

```
CAA-TP-

SUBROUTINE CCLIST (IG, ICRP, IND)

NAME: CCLIST (IG, ICRP, IND)

PURPOSE: THE CCLIST (CONSTRAINED COST REQUIREMENTS LIST) SUBROUTINE

PRINTS THE CONSTRAINED COST REGUUIREMENTS SOLUTION.

CALLED BY: MAIN PROGRAM

CALLS

-SUBROUTINE MAXC: ORDERS THE LIST

OUTPUT - ""
                                                                                                     COMMON
                                                                                                                                               40N
AC(12D),
ALLCWE(12D),
AVM(12D),
CEMCA(3DD),
CNCS(30D),
DCOSTF(12D),
DCC(3DD),
FHR(12D),
TFS(3DD),
                                                                                                                                                                                                                                                                                                                                             ACL,
AMSN(300),
BCY(300),
COST(300),
CCY(300),
DCY(300),
FHA(120),
ICOST,
TMSEL,
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             ALLOW1(120),
AVFV6(6),
CASE,
CMINT,
DCOST1(3D0),
CMC(3D0),
FHPAPD(3,120),
IFHC(120),
TNT,
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  ADESC (300),
ASUFY (120),
BF (300),
                                                                                          ٠
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              BF (300),
CL,
CPNCS (300),
DF (300),
FHM,
IDCC(2),
INS(300),
IRO(300),
NP2,
PNC(120),
STK(3C0),
TSUMB
                                                                                         ********

    DCOSTICIZO: CUSILIZUDI: CPNCSIGDD: CCOSTICISO:

    DCOSIDDI: FH4[2C1: FHM, FHM, FPAPD[3]2D];

    PFGF12DI: COST: ICCC1: IFHC12DI;

    IFSGDD: IFSEL: INSGDD: IFFGT2DI:

    IFSGDD: IFSEL: INSGDD: IFFGT2DI:

    PFGF13DD: IFSEL: INSGDD: IFFGT3DDI:

    PFGF713DD: IFGT3DDI: SFFGT3DDI: IFFGT3DDI:

    PFGF713DD: SFFGT12DDI: SFFGT3DDI: ISHORT,

    PFGF713DD: SFFGT12DDI: SFFGT3DDI: SFFGT3DDI:

    CHARACTERSID: IFFGT3DDI: 
    CHARACTERSID: IFFGT3DDI:

    CHARACTERSID: IFFGT3DDI:

    CHARACTERSID: IFFGT3DDI:

    CHARACTERSID:

    CHARACTERSID: IFFGT3DDI:

    CHARACTERSID: IFFGT3DDI:

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    CHARACTERSID: IFFGT3DDI:

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(NOT USED)

SUBROUTINE NCRNC

SUBROUTINE NCRNC (ND,12,120) NAME: NCRNC TYPE: SUBROUTINE PURPOSE: THE NCRNC (NO CANNIPALIZATION REQUIREMENTS) SUBPOUTINE GENERATES A LEAST COST FOMMIS MIX OF SPARE FARTS NEOED TO ACHIEVE A FLEET FLYING HR PROGRAM/AVAILABILITY OBLECTIVE USING A USER-SPECIFIED PARTS REPLACEMENT POLICY AND UNCONSTRAINED COSTS. CALLED BY: SUBROUTINE UCROPS CALLS -FUNCTION SR: COMPUTES CUPULATIVE NET DEMAND THRU A SPECIFIED DAY FOR A SPECIFIED PART FILES USED : NO FILES READ OR WRITTEN LES USED : NO FILES READ OR WRITTEN COMMON AC(12D), ACL, APENCISCI (300), ALLOWI(12D), ALLOWB(12O), ACL, APENCISCI (300), ALLOWI(12D), ALLOWB(12O), CY(130C), BF(13D0), AVAYG(64), CASE, COPTA(1300), CY(130C), BF(13D0), CASE, CONTA(1300), CY(130C), BF(13D0), DCOSTI(1300), CONSTF(12D), CCY(130C), BF(13D0), DCOSTI(1300), CONSTF(12D), CY(130C), CPNCS(1300), DCOSTI(1300), FHE(12D), CY(130C), CPNCS(12D), DPC(1300), FHE(12D), TASEL, INS(1300), INT, FF(12D), TASEL, INS(1300), INT, FF(12D), DCO, SENARI(1300), STK(13C0), SUMB(12D), MATTERS(1300), TSYK(1300), STK(13C0), SUMB(12D), MATTERS(1300), TSYK(1300), STK(13C0), SUMB(12D), CHARACTERS(160), SUMP(12D) CHARACTERS(160), SUMP(12D) CHARACTERS(160), SUMP(12D) CHARACTERS(160), SUMP(12D) CHARACTERS(160), SUMP(12D) CHARACTERS(160), SUMP(12D) SUMP(1)=D. DO 700 K=1,ND SUMP(1)=D. DO 700 K=1,ND SUMP(1)=D. DO 700 K=1,ND SUMP(1)=D. DO 700 K=1,ND SUMP(1)=SUMS(1) IF (12.LT.NA1 GO TO ACD COMCA(11)=SR(1)] IF (110.CE0.2) SUMP(1)=SUMP(1)*AMAX1(0.,(CRNCS(11)-COMDA(11))) SUMP(1)=SUMB7(1)*C(MPC)(1)*AMAX1(0.,(CRNCS(11)-COMDA(11))) SUMP2(1)=SUMB7(1)*C(MPC)(1)*AMAX1(0.,(CRNCS(11)-COMDA(11))) SUMP2(1)=SUMP3(1)*C(MPC)(1)*AMAX1(0.,(CRNCS(11)-COMDA(11))) SUMP2(1)=SUMP3(1)*C(MPC)(1)*CMP(1),D.] IF (110.CHCAS(11))*CMP3(1)*CMP(1),D.] IF (110.CHCAS(11))*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1))*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1)*CMP3(1)*CMP3 COMMON 100 TSUPB = AMAX1 (SUMB2(I) = FUMB + SUMP(I), D.) IF ((TSUMB-CRNCS(II)) + GE + ALLOW1(I)) CPNCS(II) = TSUPB - ALLOW1(I) SUMR = SUMR + CRNCS(II) GO TO TOD 400 ZINT = MIND(INT, NA - I2) IF (I2+GE + (TSUMB++S)) RETURN IL=INS(NP2-K+1) IF ((CRNCS(IL)+(IND-1)*7INT*TSTK(IL)) + GE + CDMDA(IL)) 60 TO 7DD Z=CRNCS(IL)+2INT TZ=2+(IND-1)*2INT*TSTK(IL)-CDMDA(IL) IF (TZ+LE+D) 60 TO 5DT CRNCS(IL)=CDMDA(IL)-(IND-1)*ZINT*TSTK(IL) TOTZ=TOTZ+7INT-TZ IF (TOTZ+LT+TNT) 6D TO 7DD CRNCS(IL)=CFNCS(IL)+AMINI(ZINT-TOTZ+ZINT) 600 TSUME=TSUME-ZINT RETURN 7DD CONTINUE IF (IZ+LF+NA+OR+IND+ETSUMB-CRNCS(II) IF (IZ+LT+NA+OR+IND+ED+1) RETURN DO 8DD K=1,NP2 J=INS(K) 80D CRNCS(J)=CRNCS(J)-TSTK(J) RETURN END ENO

81

(NOT USED)

SUBROUTINE UCCAP

SUBROUTINE UCCAP (IND) NAME: UCCAP TYPE: SUBROUTINE 000000000000000000 PURPOSE: THE UCCAP (UNCONSTRAINED COST CAPABILITY ASSESSMENT) SUBROUTINE COMPUTES FLEET CAPABILITY (AVC AVAILABILITY, PGM FLYING HRS/ACFT/DAY) BASE ON THE UNCONSTRAINED COST SOLUTION RGMNT BEING STOGKED IN THE WAR RESERVE CALLEO BY: MAIN PROGRAM CALLS -FUNCTION SR: COMPUTES CUMULATIVE NET DEMAND THRU A SPECIFIED DAY FOR A SPECIFIED PART FILES USED : INPUT - NONE OUTPUT - UNIT 6(PRINT) C COMMON AC(120), ALLCWB(120), CDMDA(300), CNCS(300), CNCS(300), DCOSTF(120), DO0(300), FHF(120), IFS(300), IFS(300), NP, ACL, AMSN(300), BCY(300), CF(300), COST(300), OCY(300), FH1(120), ICOST, INSEL, INSEL, IRC(300), ADESC (300), ASURV (120), BF (300), ALLOW1(120), AVAVG(6), CASE, CMINT, DCOST1(300), ++++ BF(300), CL, CRNCS(300), DF(300), FHM, IOCC(2), INS(3^0), INS(3^0), INS(3^0), NP2, PNC(120), STK(300), TSUMB + + CHO(300), CHO(300), FHFAPO(3,120), IFHC(120), INT, ___ +++ DOU 1300 /*
 If HF (120); If CCST; IOCC(2); IF HC (120);
 If S(300); IMSEL; IMS(300); ISHORT,
 MP; FP (300); SMART[300]; SMART, MP2;
 PTOF (300); SMART[300]; STK (300); SUMB (120);
 STK (300); SMART[300]; STK (300); SUMB (120);
 CHARACTFRE16
 ADSC; ADSC; AHSN; CASE;
 TAV=0;
 ISHORT. AX=1 - (ALLOWB(I)/(ASURV(I)+.000001)) IF (MCO(I-).50).NE.7) GC TO 1100 WRITE (6.1400) CASE WRITE (6.1500) IF (IND-EQ.1) WRITE (6.1600)

123456789012345678901234567890123456789012345

82	IF (IMD.E.C.2) WRITE (6,1700)
83	VEITE (6,1800)
85	WRITE (6.1800)
86	WRITE (6;2000)
87	WRITE (6,2100)
89	
90	AVAVG(1)=AVAVG(1)+6VC(1)+ASURV(1)
91	AVAVG(2)=AVAVC(2)+AX#ASURY(1) AVAVG(3)=AVAVC(3)+EVADO(1),T\#ONC(1)#ASURV(1)/(TAV1+-0001)
93	RAVE* FLYING HR PROG*
94	1F (1FHC(1).EQ.1) RAY=' AVAIL CONSTRAIN'
95	1200 WRILE (6,2200) I,RNC(I),AX,RAV,AVM(I),FHPAPO(I,I),I DO 1300 WRI-2
97	1300 AVAVGIKJEAVAVCKKJ/TSUPV
98	WRITE (6,2300) (AVAVG(K),K=1,3)
100	HELDERN 1400 FORMAT (1H1.30X.°CASE= '.A16)
101	1500 FORMAT 1/130X, ** FORTE CAPABILITY GIVEN THAT THE COMPUTED . REQ
102	+UIREMENT (FOR EACH POLICY) IS STOCKED ##*)
104	+TOCKED AT RETAIL (NO POST 0-DAY PARTS DEPLOYED) ++*)
105	1700 FORMAT 12/115X **** CASES ASSUME RESTOUALTINT STR=CURR STR)*,* RE
106	+CMIS AFE STOCKED AND DEPLOYED ###*)
108	1900 FORMAT (/,9x, "AIRCRAFT AVAILABILITY", 30x, "FLY HRS/AC /DAY")
109	2000 FORMAT (25%, "PART", 38%, "PART")
111	4 TX SUB SX STAT SAS SUB ST RED AVAIL AVAIL S SUBRE " AVAIL + AVAIL ST SUBRE " AVAIL
112	2200 FORHAT (17X,14,F8.3,6X,F5.3,A16,F5.2,F10.1,18)
113	2300 FORMAT (/,1X,* AVERAGE= *,11X,F5.3,6X,F5.3,21X,F10.1)
114	

SUBROUTINE UCROPS

```
SUBROUTIN
C NAME: UCRQPS
C PURPOSE: THE
C SUBROUTINE CC
C PARTS NEEDED
C CALLED BY: M/
C CALLS
-FUNCTION
C CALLS
-SUBROUT
-SUBROUT
C FILES USED :
C DIMENSIO
                              SUBROUTINE UCROPS (IND.IDFT4.IOPT5.IORD)
: UCROPS TYPE: SUBROUTINE
          PURPOSE: THE UCPOPS (UNCONSTRAINED COST ROMNTS-PARTIAL SUBSTITUTION)
SUBROUTINE COMPUTES AND PRINTS THE LEAST COST ROMNTS MIX OF SPAPE PARTS
PARTS NEEDED.GIVEN UNCONSTRAINED FUNDS.TO ACHIEVE THE CASE OBJECTIVE
           CALLED BY: MAIN PROGRAM
          CALLS

-FUNCTION SR: COMPUTES CUMULATIVE NET DEMAND THRU A SPECIFIED DAY

FOR A SPECIFIED PART

-SUBRCUTINE MAXC: DRDSRS LIST OF PART ROMNTS TO BE PRINTED

-SUBRCUTINE NCRNC: COMPUTES THE POMNT SOLUTION FOR THE "NO SUB" PART

SET AND A SPECIFIC ALLOWED STOCKOUT FOR THAT SET
          FILES USED : INPUT - NONE
OUTPUT - UNIT 6 (PRINT)
                              DIMENSION
                                                RMIN(300)
                          ٠
                            COMMON

AC(120)

ALLCWB(120)

ALLCWB(120)

CMFCA(300)

CMFCA(300)

CNCS(300)

DCOSTF(120)
                                                                                                                       ACL,
AMSN(30D),
ECY(30D),
CF(30D),
COST(37D),
DCY(30C),
                                                                                                                                                                                                                                                                   ALLOW1(120),
AVAVG(6),
CASE,
CMINT,
DCCST1(300),
DMC(300),
FHPAPD(3,120)
                                                                                                                                                                                              ADESC(300),
ASURV(120),
                          .
                          ٠
                                                                                                                                                                                             ASURV(120),
BF(300),
CL,
CRNCS(300),
DF(300),
                          +
                          +++
                                                                                                                                                                                             DF (300),
FHM,
IDCC(2),
INS(370),
IRO(300),
NP2,
RNC(120),
STK(300),
TSUMB
                                                DOC (300),
FHF (120),
IFS(300),
IPT(100),
                                                                                                                        FH# (120),
ICOST,
                                                                                                                                                                                                                                                                     FHPAPD(3,120),
IFHC(120),
                          ٠
                          +++
                                                                                                                       1051,
IMSEL,
IRC(300),
NP1,
QP4(300),
SRMAX1(300),
TSTK(300),
                                                                                                                                                                                                                                                                   INT,
ISHOPT,
NU,
PNCS(300),
SUMB(120),
                          ٠
                                                NP,

NP,

PTDFP(300,24),

SM(120,100),

TFNCS(300),

• PTOFP(300,24), OPA(300), F
• SM(120,100), SRMAX1(300), S
• TRNCS(3C0), TSTK(370), T
CHARACTER*16
• ADESC, ADSC, ADSC,
D0 100 K=1,IMSEL
00 100 I=1,120
100 SM(1,K)=0.
D0 20C J=1,NP
TSTK(J)=STK(J)
RNCS(J)=0.
CDMDA(J)=0.
200 SRMAX1(J)=-999.
D0 30C I=1,NW
ALL(V([I]=ALLOWP(I])
DCOST1(I]=0.
300 DCOSTF(I)=0.
300 DCOSTF(I)=0.
300 DCOSTF(I)=0.
300 DCOSTF(I)=0.
300 DCOSTF(I)=0.
300 DCOSTS(I)=1,NW
IA=(I-1)/5+1
TALLOWB(I)+1.5
D0 4CD J=1,NP
TSTK(J)=TSTK(J)*PTDEP(J,IA)/5.
400 RMIN(J)=0.
21NT=INT
IA00=0
IF (MOD(NA-1,INT).NE.C) IADD=1
MULT=(NA-1)/INT)+IA0D
NAD=PULT*INT*I
LAST=0
00 I500 L1=1,NA0,INT
L2=MIN0(L1,NA)
II=2-1
If (MOD(NA-1,INT).NE.C) IADD=1
MULT=(INA-1)/INT)+IA0D
NAD=PULT*INT*I
LAST=0
00 I500 L1=1,NP1
I]=2-1
IF (NP1.E0.0) GO TO 500
CDMO=COMDA(J)
IF (IN0.EC.2) XXX=XXX-ISTK(J)
IF (IN0.EC.2) XXX=XXX-ISTK(J)
IF (IN0.EC.2) XXX=XXX-ISTK(J)
IF (IN0.EC.2) XXX=XXX-ISTK(J)

                          .
                          .
                                                                                                                                                                                              AHSN.
                                                                                                                                                                                                                                                                     CASE
                                                           XXX=COMDA(J)

IF (IND •EC•2) XXX=XXX-TSTK(J)

IF (XXX•GE•SRMAY1(J)) SRMAX1(J)=XXX

CFNCS(J)=AMAX1(D••SPMAX1(J))

GO TO 6CD
```

82	500	IF (L1.CE.NA) ZINT=NA-LAST CENES (1)=AMAX163 - CENES (1)=ZINT+OP4(1))
84 85 86	600 700	CRNCS(J)=AMAX1(CRNCS(J),RNCS(J)) IF (NP2.GT.D) CALL NCPNC (I,I2,IND) TDTC=D.
87 88 89 90	800	00 P00 J=1,NP ToTC=TOTC+COST(J)+CPNCS(J) IF (TOTC-CE-CM1NT) 60 TO 1000 TA+10N=T2
91 92 93 94	900 1000	CHINT=TOTC D0 900 J=1,NP RmIn(J)=CRNCS(J) IF (L2-NE-NA-AND-NP1-NE-D) 60 TO 1400
95 96 97 98		00 1200 J=1,NP PICS(J)=AHAY1(FMIN(J),RNCS(J)) DCOSTF(I)=DCOSTF(I)+RNCS(J)+COST(J) CO_1100 M=1,IMSEL
99 100 101 102	1100 1200	IF (J.EQ.IPT(")) SM(I,M)=RNCS(J) CONTINUE CNCS(J)=CCST(J)*RNCS(J) IF (NP1-EQ.D) GO TO 1600
104 105 106 107	1300 1400 1500	II=IFS(J) DCCST1(I)=DCOST1(I)+RNCS(II)+COST(II) IF (NP1+EC+D) 60 TO 1600 LAST=11
108	1600	ALEONICIJ=TALLOW CONTINUE
110 111 112 113 114		IF (ICOST.EC.1) RETURN WRITE (6,28CO) CASE IF (INC.EC.1) WRITE (6,2900) IF (INC.EC.2) WRITE (6,3000) IF (ICOST.EC.1) WRITE (6,3100) CL,ACL,IDCC(IND) WRITE (6.3200)
116 117 118 119 120		WRÎTÊ (6,33CÔ) CMINT IF (ICCST.EC.1) RÊTURN IF (ICRC.LE.D) GO TO 1900 IF (ICCST.EO.1) RETURN IF (ICFT4.LE.D.AND.ICOST.EO.D) GO TO 2200
122	1700	
124	1.00	NOUMMY=NP DO 18CO K=1.NP
126		CALL PARC INDUNTY NOUTI
129	1800	000(II)=-1. 002(II)=-1.
131		II=IFO(I) IF (IORD_LE_D) II=IRC(I)
133		IF (MOD(I-1,50) NE . 1) GC TO 2000 WRITE (6,2800) CASE
135		IF (IND+EC+1) WRITE (6+3400) IF (IND+E0+2) WRITE (6+3500)
137		IF (ICOST.EQ.1) WRITE (6,3100) CL,ACL,IOCC(IND)
140		WRITE (6,36CD)
142	2000	WRITE (6,3600) TC=100.+CNCS(II)/(CMINT+.000001)
144	2100 2200	WRITE (6,3900) I,II,AMSN(TI),ADESC(II),RNCS(II),CNCS(II),TC IF (ICOST.EQ.1) RETURN
146		ICOST=1 IF (IOPT5.LE.D) 60 TO 2500
149		NN=IMSEL/5 IF (HOC(IMSEL,5).NE.D) NN=NN+1
151		DO 2400 L=1,NN
153		H2=TFT(2+(1-1)+5) H3=IFT(3+(L-1)+5)
155		MÁ=IPT(Á+(Ĺ+ĺ)+5) M5=IPT(5+(Ĺ+1)+5)
132		00 2300 I=1.NV If (M00 (I-1.50).NE.D) GO TO 2300
159		WRITE (6,2800) CASE IF (IND.EQ.1) WRITE (6,4000)
162		IF (IND+EQ+2) WRITE (6,4100) WRITE (6,3600) WRITE (6,3600)
103		##11C 60974UUF F19F29F39R99F3
vPIE (6,3600) white (6,4300) Addisc(P1),Addisc(P1),Addisc(P4)

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(NOT USED)

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SUBROUTINE MAXC

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SUBROUTINE MAXC (NOUMAY, NOUT)
C NAME: MAXC TYPE: SUBROUTINE
  PURPOSE: THE MAXC SUBROUTINE FINDS THE SUBSCRIPT OF THE LARGEST (IN VALUE MEMBER OF AN ARRAY (DOD(J))
         CALLED BY:
                                               PAIN PROGRAM
SUBROUTINE UCR2PS
SUBROUTINE CCLIST
                                          -
                                          -
         CALLS : NONE
          FILES USED : NO FILES READ OR WRITTEN
                       COMMON

AC (120),

ALLGWE(120),

COMDA(300),

CNCS(300),

DCCSTF(120),

DCCSTF(120),

DCC(300),

FHR(120),

IFS(300),

IFS(300),

IFT(100),

NP,
                                                                                          ACL,

AMSN(3DD),

BCY(3DD),

CF(3DD),

CCST(3DD),

CCST(3DD),

FHF(12D),

ICOST,

IMSEL,

IRC(3DC),

NPI,

SFWAX1(3DD),

TSTK(3DD),
                                                                                                                                                                                                    ALLOW1(120),
AVAVG(6),
CASE,
CMINT,
CCOST1(300),
DMO(300),
FHPAP0(3,120),
TCUC(120)
                                                                                                                                                ADESC(300),
ASURV(120),
BF(300),
                     + + +
                                                                                                                                               BF(300),
CL,CRNCS(300),
OF(300),
FHM,
IOCC(2),
INS(300),
IRS(300),
NP2,
RNC(300),
STK(300),
TSUMB
                     •
CHARACTER*16

DO 100 J

CHARACTER*16

* ADESC, ADSC,

SMAX=2

JMAX=J

CONTINUE

* CONTINUE

* CHARACTER*16

* ADESC, ADSC,

SMAX=1

DO 100 J=1,NDUMMY

X=000 (J)

CHARACTER*16

* ADESC, ADSC,

SMAX=1

DO 100 J=1,NDUMMY

X=000 (J)

ZMAX=AMAX1(SMAX,X)

IF (ZMAX.LE.SMAX) GD TO 100

JMAX=J

SMAX=ZMAX

100 CONTINUE

NOUT=JMAX

RETURN

END
                                                                                                                                                                                                     IFHC(12D),
INT,
ISHORT,
                                                                                                                                                                                                     NW,
RNCS(300),
SUMB(120),
                                                                                                                                                AMSN,
                                                                                                                                                                                                     CASE
```

1234 56789011234 56789012345678901234567890123456789012345678901234567890123456789012345678901234567890

(NOT USED)

.

FUNCTION SR

```
FUNCTION SR (I,J,CDMD)
NAME: SR TYPE: FUNCTION
PURPOSE: THE SR (STOCK REPUIRED) FUNCTION CALCULATES THE CULMULATIVE

NET DEMAND THRU & SPECIFDIED DAY FOR & SPECIFIED PART BASED

ON A SPECIFIED FLYING PROGRAM. INITIAL INVENTORY =D IS

ASSUMED IN THIS CALCULATION. NET DEMANDS IS PASICALLY

FAILED ITEMS OFFSET PY RETURNING REPARTS. IN A SENSE IT'S THE

NET NR OF 'HOLFS' ICAUSED BY THE ITEM' WHICH AFE PRESENT ON

A SPECIFIED DAY ,ASSUMING A ZERO INITIAL INVENTORY.
       CALLED BY:

- SUBROUTINE UCROPS

- SUBROUTINE UCCAP

- SUBROUTINE CCCAP
        CALLS : NONE
       FILES USED : NO FILES READ OR WRITTEN
 CC
                        COMMON

AC(120), ACL,

ALLCWB(120), AMSN(3°0),

AVM(120), BCY(300),

COMPA(300), CF(300),

CNCS(300), COST(300),

CNCS(300), CAST(300),

COCSTF(120), OCY(300),

FHA(120), ICOST,

IFS(1300), IMSFL,

IFS(1300), IMSFL,

IFS(1300), IMSFL,

PTOEP(300,24), OPA(100),

NP,

PTOEP(300,24), OPA(100),

SM(120,100), SRMAX1(300),

CHARACTER+16

ADESC, ADSC,

ID=I-OCY(J)

IR=I-BCY(J)

ORR=0.

BPR=0.

BF(10.GT.C) DRR=DF(J)+FHA(I0)

IF (IB.GT.0) BRR=BF(J)+FHA(I0)

SR=COMD+CF(J)+FHA(I)-DRR-BRR

RETURN
                                                                                                                                                                                 ADESC(300),
ASURV(120),
BF(300),
CL,
CRNCS(300),
F(300),
FHM,
IOCC(2),
INS(300),
INS(300),
INS(300),
STK(300),
TSUMB
                                                                                                                                                                                                                                                     ALLOW1(120),
                       AVAVG(6),
CASE,
CHINT,
DCOSTI(300),
                       ٠
                       +
                                                                                                                                                                                                                                                     0M0(300),
FHFAP0(3,120),
                       IFHC(120),
INT,
ISHORT,
NW,
RNCS(300),
SUMB(120),
                       +++
                       ++
                       +++
                       +
                       ٠
                                                                                                                                                                                   ANSN .
                                                                                                                                                                                                                                                     CASE
```

(NOT USED)

.

.

•

```
SUBROUTINE DIST (IFOAY, ILDAY, DAMT, K)
C NAME: DIST
C PURPOSE: THE DIST (PARTS DISTRIBUTION)
C STARTING SPARES STOCK OF A PART TYPE OVER
C CALLED BY: MAIN PROGRAM
C CALLS : NONE
C FILES USED : NO FT
                     PURPOSE: THE DIST (PARTS DISTRIBUTION) SUBROUTINE DISTRIBUTES THE
STARTING SPARES STOCK OF A PART TYPE OVER A SERIES OF 5-DAY INTERVALS
             C

COMMON

AC(120), ACL, ADESC(300), ALLOVI(

ALLOVB(120), ACL, ADESC(300), ALLOVI(

ALLOVB(120), ACL, ADESC(300), CASE,

COMOA(300), CF(300), BF(300), CASE,

COMOA(300), CF(300), CL, CHINT,

CCCS(300), COST(100), CF(300), DF(100), DCOSTI(

DCOSTF(120), DCY(300), DF(300), DM0(300)

CO00(300), FHA(120), DF(300), INT,

FHR(120), ICOST, IOST, INS(300), INT,

FHR(120), IASEL, INS(300), INT,

PT(100), INS(300), INT,

NP1, NP2, NW1, NP2, NW1,

PTOEP(300,24), OPA(300), RNC(120), RNCS(30),

CHARACTER*16

ADSC, ADSC, AMSN, CASE

D1=-(IFDAY-1)/5)*5*IFOAY-1

DL=-(IFDAY-1)/5)*5*ILOAY

ILEMIN0(24,(IFDAY-1)/5+1)

IL=MIN0(24,(IFDAY-1)/5+1)

IL=MIN0(24,(IFDAY-1)/5+1)

IL=MIN0(24,(IFDAY-1)/5+1)

ICCTIL IND FTOEP(K,II)=PTOEP(K,I)+(S.-DI)=DAMT

IF (1.CCTIL) PTOEP(K,II)=PTOEP(K,I)=PTOEP(K,I)+5.*OAMT

IF (1.CCTIL) PTOEP(K,II)=PTOEP(K,I)=PTOEP(K,I)+5.*OAMT

RETURN

D0 CONTINUE

RETURN

END
                                                                                                                                                                                                                                                                                                                                                                                   ALLOV1(120),
AVAVG(6),
CASE,
CHINT,
DCOST1(3DD),
DMD(3DD),
FHPAPD(3,120),
IFHC(120),
INT.
                                                                                                                                                                                                                                                                                                                                                                                     NH,
RNČS(3DU),
SUMB(12D),
```

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APPENDIX C

REFERENCES

1. Penn, Saul, et al., Aircraft Spare Stockage Methodology (Aircraft Spares) Study, CAA-SR-84-12, US Army Concepts Analysis Agency, April 1984 (UNCLASSIFIED)

2. Bauman, Walter J., PARCOM User's Guide, CAA-D-84-10, US Army Concepts Analysis Agency, November 1984 (UNCLASSIFIED)

3. Bauman, Walter J., PARCOM Functional Description, CAA-D-84-15, US Army Concepts Analysis Agency, November 1984 (UNCLASSIFIED)

4. Steinhagen, Carl, et al., Maximizing Daily Helicopter Flying Hours Study (MAX FLY Study), CAA-SR-83-11, US Army Concepts Analysis Agency, August 1983 (SECRET)

GLOSSARY

AC	aircraft
acft	aircraft
AFH	achieved flying hours
ASL	authorized stockage list
AVAIL	availability
CAA	US Army Concepts Analysis Agency
CUM	cumulative
DC	depot condemnation rate
DCSLOG	Deputy Chief of Staff for Logistics
DEPL	deployed
DRT	depot repair time
EFH	estimated flying hours
EST	estimated
FHP	flying hour program
FS	full substitution
HR	hour
INIT	initial inventory
M	million
MAX	maximum
MAX FLY	Maximizing Daily Helicopter Flying Hours (study)
MIN	minimum
NMCS	not mission capable due to supply
NRTS	not repairable at this station
NS	no substitution
NSN	national stock number

OPTP	Overview/PARCOM Turnkey Project
ORIG	original initial inventory
OST	order and ship time
PARCOM	Parts Requirements and Cost Model
PGM	program
PLL	prescribed load list
PT	part
QPA	quantity per application
RET	returned
RQMT	requirement
RQR	required
RT	rate
SOL	solution
STK	cumulative stock distributed
SURV	surviving
USAVSCOM	US Army Aviation Systems Command



PARTIAL SUBSTITUTION AND OTHER MODIFICATIONS TO THE PARCOM MODEL (PARCOM PARTIAL SUBSTITUTION)

STUDY SUMMARY CAA-TP-84-11

THE REASON FOR PERFORMING THE STUDY was that the two models recommended by the Aircraft Spares Study, Overview and PARCOM, could treat a fullsubstitution or a no-substitution part replacement policy but lacked the ability to represent a more realistic partial-substitution replacement policy. Of the two models, PARCOM was judged to be the better candidate for incorporation of a partial-substitution capability.

THE PRINCIPAL FINDINGS of the work reported herein are as follows:

(1) The basic PARCOM (Parts Requirements and Cost Model), developed for the Aircraft Spares Study, was extended to include the effects of partial-substitution replacement policies and deployment of initial stocks over time.

(2) The resulting extended model relates spare requirements to a flying hour/aircraft availability objective, parts replacement policy, and stockage deployment schedule--all subject to optional cost constraints. Example applications illustrated the plausibility of the model logic.

(3) The extended PARCOM significantly expands the range of application and results of the basic PARCOM methodology. As such, its implementation, in place of basic PARCOM, is warranted.

THE MAIN ASSUMPTION was that partial substitution can be usefully defined in terms of a partition of part types into a full-substitution part set and a no-substitution set.

THE PRINCIPAL LIMITATION was that definitions of partial substitution other than the assumed definition might not be addressable by the extended PARCOM.

THE SCOPE OF THE STUDY addressed the relationship of spare requirements and fleet capability for a notional Army aviation program to a flying hour/availability objective, part replacement (substitution) policy, and stockage deployment schedule--all subject to optional cost constraints. The study applied the subject model to an example, using four part types over 5 days, and to an all-up case, treating an AH-1S scenario involving 334 part types over 120 days.

THE STUDY OBJECTIVES were:

(1) To evaluate the potential for extending the capability of the basic PARCOM, developed in the Aircraft Spares Study, to include partial substitution and other desirable features lacking in the basic PARCOM.

(2) To make the above extensions and to report on and illustrate the application of the extended PARCOM and methodology.

THE BASIC APPROACH was:

(1) To assess the limitations of the basic PARCOM.

(2) To select features and capabilities, to include partial substitution, for incorporation into an extended PARCOM.

(3) To develop an extended PARCOM incorporating the selected capabilities.

(4) To report on the nature of the extended PARCOM methodology and model through exposition and illustrative example applications.

THE STUDY SPONSOR was the Deputy Chief of Staff for Logistics, Headquarters, Department of the Army.

THE STUDY EFFORT was conducted by Mr. Walter J. Bauman, Force Systems Directorate, US Army Concepts Analysis Agency.

<u>COMMENTS AND QUESTIONS</u> may be directed to the Director, US Army Concepts Analysis Agency, ATTN: CSCA-FS, 8120 Woodmont Avenue, Bethesda, Maryland 20814-2797.



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