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NAVAL POSTGRADUATE SCHOOL Monterey, Californía

Commodore Robert H. Shumaker Superintendent David A. Schrady Provost

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On the basis of sustained performance, six coatings were selected for heat-transfer performance evaluation. A separate apparatus was used in which coated copper tubes were mounted horizontally in a test section through which steam flowed vertically downward. Vapor-side heat-transfer coefficients were inferred from overall measurements. Test results indicate that the outside condensing coefficient can be increased by a factor of five to eight through the use of polymer coatings to promote dropwise condensation. Approved for public release: distribution unliminat.

An Evaluation of Polymer Coatings for the Promotion of Dropwise Condensation of Steam

by

Kenneth M. Holden II Lieutenant Commander, NOAA E.S.M.E., University of Texas, Austin, 1970

Submitted in partial fulfillment of the requirements for the degrees of

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#### ABSTRACT

Fifteen polymer coatings were evaluated for their ability to promote and sustain dropwise condensation of steam to enhance the heat transfer capability of steam condensers. Of the fifteen coatings, nine employed a fluoropolymer as a major constituent. Of the other six, four were hydrocarbons, one a chlorocarbon and one a silicone. Each coating was applied to four different metal substrates: brass, copper, naval brass, and titanium. While exposed to steam at atmospheric pressure, each coating was visually evaluated for its ability to promote dropwise condensation. Cbservations were conducted over a period of 4000 hours. Hardness and adhesion tests were performed on selected specimens both before and after exposure.

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#### BOMENCLATURE

Inside area  $l(D_i)\pi$ ,  $m^2$ A . Outside area  $l(D_0) \pi$ ,  $m^2$ A o AD As delivered BIASS Er Specific heat, kJ/kg K c<sub>n</sub> Corper Cu Inner diameter, m E i Outer diameter, m Do Inside heat-transfer coefficient, W/m<sup>2</sup>K h i Outside heat-transfer coefficient, W/m<sup>2</sup>K h Thermal conductivity, W/m K k Thermal conductivity of the liquid, W/m K k 1 1 Length of tube, m E Mass flow rate of cooling water, kg/s NBr Naval brass Fr Prandtl number C Heat-transfer rate, W Heat flux, W/m<sup>2</sup> g Rough surface finish E Reynolds number Rə Wall thermal resistance,  $m^2K/W$ **BW** Smooth surface finish 2 Temperature, Т K Ccclant inlet temperature, K 1<sub>i</sub> Coclant outlet temperature, K Т Titanium Ii  $\Delta T$ Coclant Temperature rise, K LMID,  $(T/ln((T_{sat}-T_{ci})/(T_{sat}-T_{co})), K$ Ilm U Unknown or progrietary U°. Overall heat-transfer coefficient, W/m<sup>2</sup>K

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## GREEK SYMBOLS

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- ∆ Differential
- $\mu$  Dynamic viscosity, N s/m<sup>2</sup>
- v Kinematic viscosity,  $m^2/s$
- σ Surface Tension, dynes/cm

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#### I. INTRODUCTION

#### A. EACKGROUND INFORMATION

The realization that the earth's resources are finite and not equally distributed has been amply demonstrated by the dramatic rise in the cost of energy and raw materials within the past ten years. In addition, the cost of construction for the basic Rankine cycle power plant, whether marine or stationary, has also increased. These facts have pointed to the need to apply advanced technology to beilers and condensers.

Marine applications pose the greatest problems since the power plant is limited by both size and cost. In the past, most of the effort in increasing efficiency has been directed at boiler design, while marine condensers have remained basically unchanged. For naval applications, 'the increased use of gas turbine and diesel engines for main propulsion has diminished the importance of the improvements achieved in fossil fueled boiler design. However, with the introduction of the Rankine Cycle Energy Recovery System (RACER) and the existence of numerous nuclear propulsion systems, the time has come to direct attention to the improvement of marine condensers.

Search [Ref. 1] performed a feasibility study to determine what improvements could be made in marine condensers. His research indicates that an increase of more than thirty percent in heat transfer, a thirty percent reduction in weight, and a twenty per cent reduction in volume could all be achieved if the condensation mode within the condenser were dropwise rather than filmwise. Unfortunately, filmwise condensation is the normal, stable mode of condensation on virtually all condenser materials and staps must therefore be taken to promote the dropwise mode.

## E. FILMWISE VS. DROPWISE CONDENSATION

Filmwise condensation is characterized by the formation cf a continuous sheet of liquid on the surface of the Because of the relatively simple shape and flow condenser. of the condensate film, the heat-transfer rate can be analytically predicted using the Nusselt analysis. This sheet of liquid forms a relatively high conduction resistance between the condensing vapor and the condenser surface. Under most conditions, it is this resistance which limits the heat transfer rate. For steam, the external heat transfer coefficient associated with filowise condensation is normally in the range of 10,000 to 14,000  $W/m^2$ K. On the cther hand, dropwise condensation can produce heat transfer coefficients up to twenty times that of filmwise condensa-This marked increase is a result of the process of tion. drop formation and removal from the condenser surface.

In 1939, the theory that primary drops formed at submicroscopic nucleation sites on the surface Vas fizst This nucleation theory of drop initiation was proposed. later elaborated upon by McCormick and Westwater [Ref. 10] and confirmed by Reisbig [Ref. 11] through the use of microphotography. It is during this phase of drop formation that the major proportion of the heat transfer takes place. As the drop grows, it comes in contact with many other small drops which coalesce to form larger drops. This coalescence uncovers nucleation sites which immediately start to form additional drops. When the drop is approximately 0.15 mm in diameter, the increase in conduction resistance greatly reduces direct condensation. Experimental evidence indicates that ever 90% of the heat transfer takes place during

the nucleation and growth phase. The drop continues of grow, primarily by coalescence, until it reaches a critical size at which point external forces, such as gravity of vapor shear, overcome the surface tension of the drop with the condenser surface. As the drop departs a vertical surface, it coalesces with drops in its path, wiping them from the surface and exposing additional nucleation sites.

Graham [Ref. 2] has shown that the condenser surface during dropwise condensation can be characterized as follows: 10% bare surface, 30% active drops (less than 0.15 sm in diameter) and 60% dead drops (greater than 0.15 mm in diameter). It is important to note that vapor condenses cnly on the surface of the drops and not on the bare surface areas. In addition, very little heat is transferred through the dead drops. Therefore, at any point in time, cnly 30% of the condenser surface is transferring over 90% of the heat. The large heat-transfer rates associated with dropwise condensation are due to the fact that the average condensate thickness is much less then that associated with filmwise condensation, thus greatly reducing the conduction resistance. Figure 1.1 [Ref. 14] depicts the relative sizes of the various stages and modes of condensation.

#### C. DRCP FORMATION

In 1804 Thomas Young first proposed the following equation to describe the equilibrium condition of a drop on a solid surface:

# $\sigma_{\rm sy} - \sigma_{\rm sl} = \sigma_{\rm ly} \cos \Theta \tag{1.1}$

where  $\sigma_{sv}$  is the surface tension at the solid-vapor interface,  $\sigma_{s1}$  the surface tension at the solid liquid interface,  $\sigma_{1v}$  the surface tension at the liquid vapor interface and



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Figure 1.1 Relative Sizes of Condensation Modes.

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is defined as the contact angle (Figure 1.2). mile. Θ equation is deceptively simple and poses many experimental and conceptual problems. However, the notion of a contact angle to characterize the degree to which a surface will wet is extremely useful. Zisman [Ref. 13] and other investigators have shown that there is a rectilinear relationship between  $\sigma_{1,v}$  and the contact angle. Extrapolating the data to the point at which cosine 0 =1.0, that point at which the surface completely wets, it is possible to define an empirical value, the critical surface tension, to characterize the wettability of low energy solid surfaces. Table I lists the critical surface tensions for various low-energy organic It follows from the definition of the critical surfaces. surface tension, that a liquid with a surface tension greater than the critical surface tension of the surface will rct spread. In addition, the results of many experiments indicate that the wettability of high energy surfaces coated by organic films are essentially determined by the nature and packing of the exposed surface atoms. Thus, a high energy surface can be made hydrophobic if it can be coated with a substance which has a critical surface tension less than that of the condensing liquid. In the case of water, which has a liquid-vapor surface tension of 71.9 dynes/cm at 25 C, all of the surfaces listed in Table I will te hydrophobic with the heavily fluorinated surfaces the most hydrophobic [Ref. 13].

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#### D. FACTORS WHICH INFLUENCE DROPWISE CONDENSATION

Since metal surfaces have a relatively high surface energy, the most important factor is the existence of a low energy barrier or promoter on the condenser such that the critical surface tension of the condenser surface is less than that of the cordensing liquid. This is a necessary

TABLE I Critical Surface Tensions of Low Energy Surfaces Surface Constitution dynes/cm at 20 C A. Fluorocarbon Surfaces -CF3 ô -CF2H 15  $-CF_3$  and  $-CF_2-$ 17 18 -CF2-20 -CH2-CF3 -CF2-CFH-22 -CF2-CH2-25 28 -CFH-CH2-B. Hydrocarbon Surfaces 22 -CH<sub>3</sub> (crystal) -CH<sub>3</sub> (monolayer) 24 31 : -CH2- $-CH_2 - and --CH_{++}$ 33 --CH-- (phenyl ring edge) 35 C. Chlorocarbon Surfaces 39 -CC1H-CH2--CC12-CH2-40 ==CC1<sub>2</sub> 43 D. Nitrated Hydrocarbon Surfaces -CH<sub>2</sub>ONO<sub>2</sub> (crystal) 40 -C(NO<sub>2</sub>)<sub>3</sub> (monolayer) 42 -CH<sub>2</sub>NHNO<sub>2</sub> (crystal) 44 -CH<sub>2</sub>ONO<sub>2</sub> (crystal) 45

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condition for dropwise condensation. Tanasawa [Ref. 3] noted an additional twenty-one factors which have, no a greater or lasser degree, an influence upon the heattransfer coefficient of dropwise condensation. Of these additional factors, the most important are: (1) the thermal properties of the condenser material, (2) non-condensing gases, (3) external forces, and (4) condenser surface roughness.

The effect of the thermal properties from which the condenser is manufactured on the heat transfer coefficient of dropwise condensation is not completely resolved. Hannewan and Mikic [Ref. 4] have proposed the theory that the non-uniformity of drop size and spacing produces a nonuniformity in the surface wall temperature since the large drops act as insulators while the small drops are highly conductive. The net result is a thermal constriction resis-Materials of low thermal conductivity allow less tance. lateral heat transfer at the surface and therefore have a high constriction resistance. Rose [Ref. 5] believes that the non-uniformities are rapidly homogenized by the frequent coalescence between drops. Therefore, the constriction resistance will be small. Data are available to support both points of view. Regardless of whether the constriction resistance is large cr small, the material selected for condenser construction will be limited by other constraints such as durability, machineability, availability and cost.

The effects of the second and third factors, noncondensing gases and external forces, can be minimized by proper design. The condenser must be able to operate under vacuum conditions without producing stagnant regions within the hundle where non-condensing gases can concentrate. The design should be such that gravity and vapor shear work together to remove the condensate drops.

The forth factor is surface roughness. For liquids exhibiting contact angles less than 90 degrees, a cruck surface decreases the contact angle. This produces pecter quality condensation by causing the larger drops to occupy a greater surface area than would be normal. There is less room for active drops and more potential nucleation sites are covered. Both Griffith [Ref. 15] and Graham [Ref. 14] reported reduced heat transfer as a result of surface roughness. However, as will be pointed out later in this thesis, a certain amount of surface roughness is necessary for the proper adhesion of some of the organic polymers tested.

#### E. FECMCTICN OF PERMANENT DROPWISE CONDENSATION

Dropwise condensation can be promoted by: (1) direct application of a suitable chemical, such as oleic acid or montan wax, to the condenser surface, (2) injecting nonwetting chemicals into the wapor which adsorb on the condenser surface, and (3) using a low energy polymer or noble metal coating.

Both the first and second method are limited in that neither produces a permanent hydrophobic coating. Results wary widely but usually the effectiveness of these promoters is measured in the hundreds of hours. The third method, the use of permanent coatings, has been the subject of considerable interest. At present, there are two types of coatings which can be used to promote permanent dropwise condensation: noble metals and organic polymers.

## 1. Noble Metals

Cf the noble metals, only gold has been shown to consistently produce excellent dropwise condensation. This would appear to contradict theory since the surface energy cf gold is relatively high. Recent experiments have shown

that a "clean" gold surface applied under carefully controlled conditions will spontaneously wet as predicted. However, gold has the ability to attract and retain organics which render the surface hydrophobic. For this reason, gold is referred to as a "self-promoter." An extensive investigation by Woodruff [Bef. 12] concluded that the promotion of dropwise condensation of a gold-plated surface was directly related to the surface gold and carbon concentrations and inversely related to the copper, aluminum and oxygen concentrations.

Although gold has been shown to produce excellent dropwise condensation, its behavior over extended periods under varying conditions is uncertain. In addition, practical considerations such as cost and availability make gold unattractive for this purpose.

## 2. Crganic Polymers

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The second rethod of producing permanent dropwise condensation is to apply an organic polymer with a low surface energy to the condensing surface. Although many experiments have beer conducted, virtually all of the reasearch has been done using a fluorocarbon or silicone polymer.

Flucrocarbon polymers exhibit the desirable characteristics of high thermal stability, very low surface energy, and are highly resistant to chemical attack. However, they also have the undesirable characteristic of a relatively low thermal conductivity usually on the order of 0.3 W/m K. Assuming a realistic value of a tenfold increase in the outside heat-transfer coefficient due to dropwise rather than filmwise condensation and a nominal value of 0.3 W/m K for thermal conductivity, the maximum film thickness which could be employed to produce an increase in the overall heat-transfer coefficient would be on the order of

20 micrometers. Films of greater thickness would impose a thermal harrier so large that it would negate any batefic derived from altering the mode of condensation. The most common polymer which has been employed is PTFE (polytetrafluorcethylene). PTFE is commonly referred to as Teflon, EuPont's registered trademark for a non-stick finish.

In 1956, Smith [Ref. 7] concluded that, when applied hy utilizing the existing application techniques, thin Tefler coatings were not satisfactory for increasing overall heat-transfer rates for Naval applications. In 1964, Fox [Ref. 21] reported on tests conducted with a small condenser in which the tubes had been coated by Teflon to a thickness cf 12.5 micrometers (0.0005 in). Performance was not significantly increased. Shortly thereafter, further studies into dropwise condensation by the Navy were In 1966, Brown and Thomas [Ref. 24] conducted suspended. experiments with tubes of Admiralty brass coated with 2.5 micrometers (0.0001 inch) of PTFE. The outside heattransfer coefficient for dropwise condensation averaged ' approximately three times that of filmwise condensation. Th 1969, Graham [Ref. 14] also achieved a threefold increase using flat, copper plates coated with a Teflon film estimated to be 1.5 micrometers (0.00006 inches) thick. This represented the practical limit for the application of Teflon films using standard techniques. In 1979, both Manvel [Ref. 8] and Perkins [Bef. 22] used vacuum-deposition sputtering to apply ultra-thin layers of PTFE. Thicknesses ranged from a minimum of 0.04 micrometers to a maximum of C.13 micrometers. The results were disappointing and unexrected. Although the dropwise performance was good, the steam-side heat- transfer coefficient was increased by a maximum cf only 1.6. This result was attributed to the presence of non-condensing gases. In addition, all of the coatings showed signs of physical deterioration after very short operating periods.

The use of silicones to promote drobwise condensation began in the 1950's and continued into the early 1960's. In 1959, Bullberg and Kandall [Ref. 25] exparimented with a commercially-available, semi-permanent silicone release agent, R-671. With a film thickness estimated to be 10 micrometers (0.0004 in), an increase of 1.5 in the cverall heat-transfer coefficient was obtained. Preliminary endurance tests indicated that the coating was carable of producing dropwise condensation for periods in excess of 1500 A subsequent study conducted by hours. the Westinghouse Corporation for the U. S. Navy investigated five silicon compounds, including R-671, and Teflon. These tests produced only accerate increases in the overall heattransfer coefficient. Most important, all of the silicone coatings exhibited limited endurance. The best silicone compound had been reduced to 80% dropwise and 20% filmwise Withir 135 hours of operation. From these tests, it was concluded that Teflon was the only coating of the group that exhibited а life expectancy sufficient to warrant investigation.

1966, Ert and Thalen [Ref. 23] conducted In an extensive investigation of permanent hydrophobic coatings. Although they were primarily concerned with sulfide films and noble metals, they also investigated organic polymers. One of the polymers investigated was PTFE and another was a relatively new compound developed by the Union Carbide Corporation, parylene-N. Farylene-N is a polymer of paraxylylene which can be vapor deposited in very thin films. Unlike FTFE, parylene-N contains no fluorine and therefore would not be expected to be as hydrophobic as the fluoropolymers. However, a cne-micrometer- thick film produced good dropwise condensation in excess of 2400 hours of continuous The overall heat transfer coefficient was oparation. increased by a factor of approximately 1.5.

Recent developments in techniques for the application of ultra-thin coatings, in addition to the development of new materials, has put the use of organic polymete to promote dropwise condensation within the realm of feasibility.

## F. PURPOSE OF STUDY

The overall purpose of this study was to evaluate the performance of various organic polymers as effective dropwise promoters. This was accomplished in two distinct stages.

## 1. Indurance

Fifteen different coatings, which were applied to four different substrates with two different surface preparations, were subjected to an endurance test. The test consisted of continuous exposure to steam at atmospheric pressure. Visual observations were made to determine which coatings could sustain dropwise condensation. Those coatings which showed little or no degradation after a minimum of 2000 hours of exposure were used in the second stage of this study.

## 2. <u>Eest-Transfer Evaluation</u>

Cn the basis of the endurance tests, six coatings were selected for quantitative evaluation: (1) Nedox, a coating system developed by the General Magnaplate Corporation, (2) No-Stik, a coating developed by Plasma Ccatings, Inc., (3) parylene-N, (4) parylene-D, (5) NRL C-6 fluorcepcxy, and (6) NRL fluoroacrylic. These coatings were applied to copper tubes in order to quantify their effect on the cutside heat-transfer coefficient.

#### II. EXPERIMENTAL APPARATUS

#### A. INTRODUCTION

Tests and evaluations were conducted in two stages. Luring the first stage, prepared samples of selected coatings were subjected to a rigorous endurance test prior to the second stage, heat-transfer evaluation. During the endurance test, the coatings were visually evaluated for their ability to promote and sustain dropwise condensation. Those coatings which performed satisfactorily in excess of 2000 hours were selected for the second stage, heat-transfer evaluation. During this stage, the effect of the coating on the outside heat-transfer coefficient was determined.

## E. ENDURANCE TEST FACILITY

The endurance test consisted of exposing prepared samples to steam at atmospheric pressure while mounted on a heat sink. The heat sink was necessary to provide a thermal gradient of sufficient magnitude to induce vigorous condensaticn. The endurance test was to run continuously for an indefinite period. Therefore, a principle concern was to design and construct a facility which was simple, reliable, and could be left unattended for extended periods. In addition, since photography would be used to assess the quality cf the dropwise condensation and provide a visual record of it was essential that the specimens be changes with time, displayed with an unclatructed view. These objectives were accomplished by the construction of an endurance testing apparatus consisting of three major components: (1) steam chamber, (2) heat sink, and (3) de-superheater. Figure 2.1 depicts a schematic of this system.



Figure 2.1 Endurance Test Apparatus Schematic.

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A. 1.

## 1. Steam Chamber

The steam chamber was a rectangular box with overall dimensions 0.457 m (18 in) in length, 0.127 m (5 in) in depth and 0.165 m (6.5 in) in height. The main body of the chamber, top, bottom and end plates, was constructed from 4.8 mm (0.1875 in) thick, stainless-steel plate. The side fanels were constructed from two pieces of 6.3 mm (0.25 in) thick borrosilicate glass. The inner glass was recessed into the body of the chamber while the outer glass Was inlaid in a stainless-steel frame. Twenty-four threaded, stainless steel studs were butt welded along the periphery cf each side to hold the glass parels in place. To provide for an adequate dead air space between the glass panels, a 6.3 mm (C.25 in) thick aluminum spacer was used. The entire assembly, the inner glass, spacer, and outer frame, Was held in place by bolting the outer frame to the main chamber. A Neoprene "O" ring was used to provide a seal between the inner glass and the chamber. Both sides of the chamber were identical. Figure 2.2 shows the assembled chamber in operation.

Steam was introduced through 25.4 mm (1.0 in) OD stainless steel tubing at the top center of the chamber. The steam was distributed along the centerline through a perforated stainless steel manifold running the length of the chamber. A drain in the center of the bottom of the chamber returned the condensate to the de-superheater. A horizontal tranch line located in the return line directly beneath the chamber was open to atmosphere. This prevented the chamber from becoming pressurized. In addition, a continuous jet of steam issuing from this tube indicated that the supply to the chamber was adequate.



Figure 2.2 Steam Chamber



Figure 2.3 Steam Chamber Close-up Showing Rod Heater

In order to have a clear field of view for taking pictures, it was necessary to prevent condensation on the inside of the inner glass panel. This was accomplished by heating the air space between the glass panels with roć heaters. The heaters were 0.438 m (19.0 in) long and 3.0 mm (0.125 in) in diameter with a power output of 165 Watts each. The heaters were inserted through holes drilled in the ends of the spacer frame. Two heaters were required per side; one was placed along the bottom of the air space and the other approximately half way up. All four heaters were controlled with one rheostat. The heaters are visible in Figure 2.2.

2. <u>Heat Sink</u>

The test specimens were mounted on a heat sink which was centered within the chamber. The heat sink was fabricated from two sheets of copper 0.388 m (15.25 in) long, 95 mm (3.75 in) wide and 6.35 mm (0.25 in) thick held apart along the periphery by 6.35 mm (0.25 in) square, brass bar stock. The interior was divided into three longitudinal chambers by brass baffles. Cccling water entered and exited each charber through 6.35 mm (0.25 in) OD stainless steel tubes which protruded from the ends of the heat sink. After installing the internal baffles. the entire unit was brazed along the periphery to form a single, leak-free unit. Once assembled, the large flats were lapped to provide a smooth, uniform contact surface. The heat sink was suspended in the center of the steam chamber by passing the water inlet and cutlet tubes into bulkhead fittings mounted in the end The end plates were bolted in place before tightplates. ening the compression nuts on the bulkhead fittings. The coolant, tap water making a single pass, was supplied and discharged through 6.35 mm (0.25 in) diameter nylon tubing.
The heat sink was designed to hold eighty-four 25.4 mm (1 in) square specimens. The thickness of the specimens was nominally 0.76 mm (0.03 in) or 1.5 mm (0.060 in) depending upon the substrate material and coating system. A primary concern was the ability to hold the specimens tightly against the heat sink in order to minimize contact In order to accomplish this, four equispaced resistance. stainless steel tracks, 3.2 mm (0.125 in) by 4.7 mm (0.1875 in). were fastened to each face of the heat sink with counter-sunk machine screws. The tracks were pre-drilled and tapped to receive stainless-steel clips which were held in place by set screws. The clips were designed to securely clamp each specimen against the heat-transfer surface by the upper and lower edges.

# 3. <u>De-Superheater</u>

An additional concern was the ability to provide a reliable steam supply. House steam from a central boiler was available, but the service pressure was greatly in excess of atmospheric. Therefore, when throttled to atmospheric pressure, the steam entering the chamber would be superheated by approximately 60 K. In addition, contaminants such as rust and scale were common in the system. In crder to reduce the pressure and the degree of superheat and provide a sediment trap for non-volatile contaminants, a stainless-steel tank was fabricated. The tank was 0.66 m (26 in) in length and 0.33 m (13 in) in diameter. The bottom of the tank was situated 1.82 m (72 in) below the bottom of the steam chamber. The steam supply line from the de-superheater to the chamber was 25.4 mm (1 in) OD stairless-steel tubing, while the condensate return line was 12.7 mm (0.5 in) OD stainless-steel tubing. The combination of elevation and steam line diameters made a gravity return possible, thus eliminating the need for a condensate return pump.

Cperation commenced by injecting steam into the bottom of the de-superheater through a sparger tube. The amount of steam was controlled by a throttling valve. The steam passed through the de-superheater to the chamber, condensed on the heat sink and the condensate returned to the de-superheater. After one hour, steady-state conditions prevailed in which the de-superheater was approximately one-third full of water and the return condensate was balanced by overflow from the de-superheater. 1

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#### C. EFAT-TRANSPER EVALUATION FACILITY

The apparatus used to determine the quantitative effect of a coating on the outside heat-transfer coefficient was described in detail by Poole [Bef. 20]. A brief description of the apparatus will be given here. A schematic of the system is shown in Figure 2.4.

Steam was generated in a 0.305 m (12 ir) diameter glass toiler using ten 4000-watt immersion heaters. The steam then ressed through a 0.305 m (12 in) to 0.152 m (6 in) reducer into a 2.44 m (8 ft) vertical section, through a 180 degree bend, and then down a 1.52 m (5 ft) vertical section before entering the stainless steel test section. The tube to be tested was mounted hcrizontally in the center of the test section behind a viewing port. Steam that did not condense on the tube passed on to an auxiliary ccil condenser. All condensate was returned to the boiler by gravity through stairless steel tubing. Operating under vacuum conditions of 0.012 MPa (1.62 psia), the test apparatus produced vapor velocities of approximately 2.0 m/s (6.5 ft/s) past the tube.

Cooling water for the tubes was provided by a centrifugal pump. A throttling valve was employed to control the flow through the tube from zero to a maximum of 0.52

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liters/s (8.3 gal/min) which produced a maximum velocity of 4.3 m/s (14.0 ft/s) through the tube. A continuous supply of tap water was used for cooling the auxiliary condensar. By throttling the flow of tap water to the auxiliary condensar, the pressure within the system could be regulated.

As previously stated, the presence of non-condensing gasses can have a relatively large effect on the heattransfer rate. Extreme care was taken to ensure that the heat transfer test apparatus was virtually leak free. As documented in [Ref. 20], the experimental apparatus was successively improved until it could be repeatedly demonstrated that the rise in pressure due to the ingress of non-condensing gasses was less than 5.0 mm Hg (0.1 psia) in twenty-four hours with a system pressure of 80.0 mm Hg (1.6 psia).

After the installation of the test tube, an air ejector was used to reduce the system pressure to roughly 100.0 mm Hg (1.9 psia). At this point, the air ejector was secured and the boiler was energized. After boiling had commenced, the air ejector was again activated for approximately 10 minutes. Over many data runs, this purging procedure was validated by the absence of any improvement in the heat-transfer by additional purging.

#### 1. Instrumentation

The power to the heaters was regulated through a silicon-controlled rectifier. This provided precise control and an accurate measure of the power being consumed. A mercury-in-glass mancmeter, calibrated in millimeters, was used to measure the internal pressure of the system. The temperature rise through the tube was measured by a dual channel Hewlett-Packard 2804A quartz thermometer. In order to provide in situ verification of the readings from the





quartz thermometer, a ten-junction thermopile was also dead to measure the temperature rise. Proper shielding of the thermopile wires eliminated the considerable of first-starty interference generated by the boiler rectifier at low voltages. The significance of this problem documented by Pocle [Ref. 20]. Throughout all of the data runs, the quartz thermometer and the thermopile agreed to within ±0.03 K. A calibrated rotameter was used to measure the coolant flow rate through the tube.

All raw data were recorded on disk by a Hewlett Fackard \$826A computer. With the exception of the manometer and rotameter, all data were interfaced through a Hewlett Fackard 3497A Data Acquisition/Control Unit. The rotameter and manometer readings were manually entered from the keyboard into an interactive data logging and reduction program.

#### D. TEST SPECIMENS

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#### 1. <u>Indurance Test</u>

Four metals used in the construction of condensers were selected for use as test specimen substrates: copper, brass (70% copper, 30% zinc), Naval brass (60% copper, 39.25% zinc, 0.75% tin), and titanium. The copper, brass and titanium specimens were 0.76 mm (0.03 in) thick, while the Naval brass was 1.52 mm (0.06 in). The sheets were sheared into 25.4 mm (1.0 in) squares. Care was exercised to ensure that the shearing operation did not warp the specimens. Frice to surface preparation, the edges were lightly dressed with a file to remove the slight burn left by the shearing process.

Two surface finishes were used during this stage of testing. A "smooth" surface finish was produced by lapping the front face on a belt sander using 180 grit silicon

carbide abrasive paper. This was followed by hand stroking the specimens on 400 grit followed by 600 grit actasive paper mounted on a glass table. When changing grit, the specimen was rotated 90 degrees and stroked in only one direction until the markings from the previous abrasive were removed. The specimen was then placed in an ultra-sonic bath containing methanol for fifteen minutes and then blown dry. For adhesion purposes, a mirror-like finish was deemed undesirable. The procedure described was designed only to provide a uniform surface free of gross irregularities.

A "rough" surface was prepared by grit blasting with size 40 glass beads with the air pressure set at 0.138 MPa (20 psig). The surface roughness produced by this procedure was determined by means of a surface profilemeter. Representative samples were found to have a surface roughness ranging from thirty to fifty microinches RMS. These specimens were also cleaned in the ultra-sonic bath as previously described.

The surface preparation for the commercial coatings that were tested was dictated by the manufacturer. In most cases, the preparation was considered to be proprietary information. Therefore, if the actual surface preparation is not known, this fact will be noted. Otherwise, the terms "smooth" and "rough" refer to the procedures described in this section.

# 2. Heat-Transfer Test

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The tubes used in the heat-transfer tests were machined from thick wall, low-exygen copper pipe. The tubes were 0.129  $\pi$  (9 in) long with an inside diameter of 12.7 mm (0.5 in) and an outside diameter of 19.0 mm (0.75 in). All of the tubes were machined at the Naval Postgraduate School and then shipped to various laboratories and commercial concerns for application of the particular coating.

#### E. ECLYMER COATINGS

Fifteen organic coating systems were considered for endurance testing. Ten of the coatings are commercially available, while the remaining five are strictly experimental. A preliminary screening procedure was necessary to identify which coatings were likely to produce dropwise condensation. The principle characteristics for consideration were: (1) critical surface energy, (2) thermal stability, (3) the ability to be applied in ultra-thin layers, and (4) method of application.

The critical surface tension is not a parameter which is normally measured. However, knowing the chemical structure of the polymer, it was possible to estimate how well the coating could be expected to perform. Knowing that water has a surface tension of approximately 71.9 dynes/om and referring to Table I, it can be seen that the flucrinated polymers could be expected to produce the best dropwise condensation, with the hydrocarbon surfaces the next best cpticn. For this reason, nine of the coating systems selected contained fluorine. All of the coating systems had maximum recommended service temperatures in excess of 150 C. In addition, all but one system was advertised as being possible to apply in thicknesses of twelve micrometers (0.0005 in) or less.

It is important to note that the endurance test was designed to be rather harsh. None of the coatings tested were formulated or developed specifically for the purpose of promoting dropwise condensation. Therefore, qualitative assessments such as "poor dropwise" or "degraded with time" should not be construed as a critical statement of a coating's ability to perform satisfactorily under the environmental conditions for which it was intended. Simply stated, such comments are meant only to indicate that a coating did

not perform satisfactorily under severe conditions for which it was not designed. After the evaluation of considerable product and research data, the following coatings ware selected for evaluation:

### 1. No-Stik

No-Stik is a thermally conducting coating, impregrated with PTFE, which is applied by a thermal or plasma spray technique. Nc-Stik was developed by Plasma Coatings, Inc. for use as a hard-wearing, corrosion-resistant mold release for the paper, food and plastics industries. The actual process is proprietary information. Thermal conductivity is enhanced by loading the spray with copper during application. No-Stik has superior hardness and the ability withstand temperatures in excess of 200 C (400 F). to Unfortunately, the coating cannot be applied in thicknesses cf less then 50 micrometers (0.002 in). If the coating were strictly an organic polymer, a thickness of this magnitude would provide a thermal resistance far greater than the reduction in thermal resistance due to dropwise condensation. Bowever, since the coating is heavily doped with copper, it's thermal conductivity is difficult to estimate. Therefore, although the thickness of the coating did not fall below the estimated payback value of 20 micrometers (0.0015 in), it was still included for evaluation.

#### 2. Nedcx

Nedcx is a commercially-available coating developed by the General Magnaplate Corporation for use as a corrosion-resistant mold release. The coating technique is a proprietary process in which a very porous, hard surface of chrome-nickel alloy is electro-deposited on the substrate surface. The pores are enlarged through a series of processes to accept the infusion of PTFE which forms a

clear, ultra-thin doating over the entire surface. Protocol control allows the coating to be applied as thin as five disconstances (0.0002 in). This was the thickness selected for testing.

# 3. Emralon 333

Emralon, a registered trademark of the Acheson Colloids Company, refers to a family of resin-based, fluorofolymer coatings. Emralon 333 is a one component blend of fluorccarbon lubricants in an organic resin binder. This coating was developed to provide dry-film lubrication and release properties for a variety of industrial purposes. In addition to a service temperature in excess of 200 C, the coating exhibits excellent resistance to abrasicn, humidity, Application is achieved with an and chemical attack. external atomizer gun using an MBC #30 nozzle. The optimum thickness for wear and abrasion resistance is advertised to be twenty-five micrometers (0.001 in). The film thickness used for endurance testing was 20 micrometers (.0008 in).

4. <u>Isonel 472</u>

Iscnel is a registered trademark of Schenectady Chemicals, Inc. Isonel 472 is a clear, thermosetting, modified polyester insulating varnish. Although not expected to perform as well as the fluorinated compounds, ease of application, availability, and low cost made investigation worth-Application was accomplished by fastening a while. substrate to a horizontal turntable, placing a small amount of the varnish in the center followed by spinning at 5000 rpm for sixty seconds. A viscosity of 0.235-0.3 kg/m s at C produced a film thickness of 5 to 10 micrometers 25 (0.0002-0.0004 in). The specimen was then cured at 150 C for two hours. Application was performed at the Na val Postgraduate School.

# 5. Isonel 31-398

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Isonel 31-398 is similar to Isonel 472. The vacaish was applied in the same manner as previously described, but curing was accomplished at 100 C for one hour.

#### 6. <u>NRL C-6 Fluorcepoxy</u>

For the past fifteen years, development of fluorinated polymers has been a continuing research effort at Naval Research Laboratory located in Washington, D.C. The C-6 fluoroepoxy was developed by Dr. James Griffith at the Naval Research Laboratory in an effort to produce a thin, tough protective coating which would also provide extremely low fluid absorption. At present, the coatings are experimental and not available to the general public. All of the NRL coatings referenced in this thesis were formulated and applied by Dr. Griffith.

C-6 fluoroepcxy is a thermosetting polymer. It is so named because of the perfluorinated, straight-chain hexyl group present on the five position of the central benzene ring (see Figure 2.5). Thermosetting compounds are



#### Figure 2.5 Chemical Structure of NRL C-6 Fluoroepoxy.

characterized by being insoluble in virtually all commercial solvents and by the lack of a melting point. When exposed

to high temperatures, thermosetting compounds decompose rather than malt. These characteristics dictate that polyrefization take place on the surface after application. If is important to note that the fluorine is not located in the vicinity of the reactive epoxy groups and therefore plays no role in the polymerization reaction. In addition, each carbor-bearing fluorine is totally fluorinated. This results in maximum stability with regard to environmental degradation.

The C-6 flucroepoxy had previously been tested by both Manvel [Ref. 8] and Ferkins [Ref. 22]. The results were disappointing because of the rapid degradation of the coating. Manvel [Ref. 8] reported that the coating appeared to dissclve after a short period of exposure to steam. Since the eroxy is a thermosetting polymer, dissolution is virtually impossible under the test conditions stated. More likely, these earlier results were caused by an inadequate bond between the substrate and the coating, allowing it to be physically removed, or by an insufficient cure which allowed the coating to remain intact but lose its hydrophobic properties. After discussing these problems with Dr. Griffith, a modified C-6 was supplied for the endurance test. The catalyst used was ethylene-diamine. Curing took place at room temperature for eight hours followed by forty-eight hours at 90 C.

#### 7. NRL BCE-7 Fluoroepoxy

The BCE-7 flucroepoxy has an chemical structure as depicted in Figure 2.6. Although similar to the C-6, there are basic differences in the group attached to the five position on the central ring. In addition, a fluoro anhydride curing agent was utilized with the following cure schedule: three days at rocm temperature, twenty-four hours at 70 C and seventy-two hours at 120 C.



Figure 2.6 Chemical Structure of NRL BCE-7 Fluorcepoxy.

# 8. NRL C-6' Flucioepoxy

The C-6. fluoroepoxy was a variation of the ECE-7 fluorcepoxy formulated by using an adduct of the C-6 fluoroepoxy as the curing agent. The curing process was the same as that of the ECE-7 coating.

# 9. NRL Fluoroacrylate

Like the flucroepoxies, the NRL fluoroacrylate is a therresetting polymer. After application, the specimens were subjected to a ritrogen purge at 50 C. Catalysis was



#### Figure 2.7 Chemical Structure of NRL Fluoroacrylate.

accomplished by exposure to long wave length, ultravialelight at 80 C for 30 minutes. The chemical arrangement of the fluorcacrylate is shown in Figure 2.7.

# 10. NRI Flucroacrylic

The NRL Fluorcacrylic was characteristic of a linear thermoplastic. The acrylic differs from the epoxies and the acrylate in that there is no three dimensional, crosslinking between the polymer chains. This makes the acrylic inherently less tough. In addition, being a thermoplastic rather than a thermosetting plastic, polymerization takes place prior to application. After polymerization, the acrylic can be dissolved in a commercial solvent and then applied. In this case, freon was used as the carrier. Curing was accomplished at room temperature by evaporation



Figure 2.8 Chemical Structure of NRL Pluoroacrylic.

cf the carrier. Unlike a thermosetting plastic, the acrylic can be removed by dissolving it with freen. The chemical structure of the flucrcacrylic is shown in Figure 2.8.

#### 11. Sputtered PTFE

FIFE was sputter deposited on the various substrates at the Lewis Research Center, National Aeronautics and Spice Administration. Briefly, sputtering is accomplished by placing a target, in this case PIFE, and the substrate to be coated in a vacuum chamber. The chamber is evacuated and then tackfilled at low pressure with an inert cas. An RF power supply is utilized to provide a negative voltage on the target which causes it to become a cold cathode electron emitter. The emitted electrons ionize atoms of the gas which are then accelerated toward the negatively charged target. The positive ions strike the target with sufficient force to dislodge atoms or molecules. These dislodged molecules move through the plasma and re-polymerize on the substrates and the exposed surfaces of the chamber. A thorough description of the glow discharge process is given Ref. 9.

This process has the ability to deposit ultra-thin coatings of uniform thickness on virtually any substrate. The process does have some drawbacks. Deposition is fasically line-of-sight and relatively slow. These characteristics make it somewhat difficult to coat a round object such as a tube. The one-inch-square sputter coated specimens had a coating thickness of approximately 0.4 micrometers which required approximately one hour of deposition time.

#### 12. Farylane-N

Farylene is a generic name for a family of thermoplastic polymers developed by the Union Carbide Company for use as a conformal, insulating coating for the electronics industry. The most basic parylene member, parylene-N, has the simple chemical structure shown in Figure 2.9. Because it is not practical to melt or extrude and because it is



#### Figure 2.9 Chemical Structure of Parylene-N.

insoluble in conventional solvents, parylene is not produced as a polymer like most thermoplastics. Instead, it is produced as a dimer. The dimer is put through a two-step heating process under vacuum. This process produces a reactive monomer vapor which, when passed over an object at room temperature, polymerizes on the surface forming a uniform coating. Farylene is tough, can be applied as thin as 0.1 micrometer ( $4.0 \times 10^{-6}$  in), has excellent chemical resistance, and is thermally stable up to 120 C. Most importantly, unlike most other vacuum deposition processes, the process is not "line-of-sight." The process will produce a uniform coating on the top, fortom, and edges, inside and out, on virtually any object.

The parylene coatings were applied by the Lawrence Livermore National Laboratory (LLNL), which is licensed by the Union Carbide Corporation. Two sets of samples were coated with parylene-N; one set with a thickness of 0.5 micrometer  $(2.0 \times 10^{-5} \text{ in})$  and the other 1.0 micrometer  $(4.0 \times 10^{-5} \text{ in})$ .

#### 13. Farylane-D

Farylene-D, a double chlorinated parylens, is applied in the same manner as parylens-N. It has the chem-



#### Figure 2.10 Chemical Structure of Parylene-D.

ical structure shown in Figure 2.10. Parylene-D was selected for evaluation because it possesses a moisturevapor transmission rate only one-sixth that of parylene-N. As before, two sets of samples with thicknesses of 0.5 and 1.0 micrometer were supplied by LLNL.

# 14. Ferco 6122

Ferco 6122 is a graphite-based, solid film lubricant developed for the nuclear industry by Power and Engineered Froducts, Co. This coating is stable to 300 C, has a coefficient of friction of 0.08, is chemically resistant, and can be applied by spray as thin as 12.0 micrometers (0.0005 in).

# 15. <u>Silicone</u>

Silicone polymers possess unusually low surface tensions. Silicones are available as oils, greases, rubbers and as thermoplastic or thermosetting polymers depending upon the monomer and polymerizing conditions. The fasic silicone structure consists of a silicon-oxygen-silicon backbone combined with organic groups. By varying the organic side chains, the properties of the polyget me be varied. For example, methyl groups have been used to promote thermal stability and water repellency in silicone fluids. An examination of Table I indicates that a critical surface tension as low as 21 dynes/om could be expected from this combination.

The silicone tested was Silgan J-500, a high quality commercial coating manufactured by the SWS Silicones, Inc., known for its durability and toughness. Application of this coating was performed at the Naval Research Laboratory. Unfortunately, the coating proved to be somewhat difficult to apply which resulted in a rather thick, uneven coating. It was decided to test the coating for its endurance and hydrophotic characteristics before attempting to develop a procedure for applying a thin, uniform layer.

#### III. EXPERIMENTAL PROCEDURES

#### A. ENDUFANCE TEST

Forceps were used to hold the specimens against the heat sink while they were clamped in place. The glass side panels were then placed in position and the nuts secured finger-The three coolant control valves were opened to tight. their maximum positions allowing water to pass through the heat sink. The steam regulating valve was opened until steam issued from the branch line beneath the chamber. AΞ the steam condensed, the condensate would return and commence filling the de-superheater. Steady-state was reached in approximately one hour when the de-superheater was roughly one-third full. At this point the returned condensate was balanced by cverflow from the de-superheater discharge. The steam regulating valve was then adjusted to provide a steady flow of steam from the branch line. This ensured that the chamber was receiving the maximum amount of steam which could be condensed.

Visual observations were conducted daily. In order to provide a permanent, visual record, photographs were taken at the following intervals: 0, 260, 790, 1500, and 2000 hours. Periodically, the system was shut down in order to remove specimens which were not performing satisfactorily. These were set aside for examination under the scanning electron microscope (SEM), and new samples were installed in their places.

#### E. HEAT-TRANSPER EVALUATION PROCEDURE

With one exception, Poole [Ref. 20] provides a levelled description of the operational procedures, data reduction, and computer programs utilized during the heat-transme evaluation stage. The one exception concerns the use of a spiral insert placed within the tube being tested in order to enhance the inside heat-transfer coefficient. The justification for enhancement of the inside coefficient can be explained by examining the manner in which the outisde heat-transfer coefficient was determined.

The total heat transfer can be computed from equation (3.1); the right-hard side being measured or known quanti-

Q=mc<sub>a</sub>∆ T

(3.1)

ties. The overall heat-transfer coefficient can now be computed using equation (3.2):

 $U = C/A \Delta T lm$ (3.2)

Since the inside of the tube was thoroughly cleaned before testing, and all of the noncondensing gases were effectively eliminated, the resistances due to these two factors were considered negligible.

Since the inside heat-transfer coefficient had previcusly been established using the Sieder-Tate correlation and the wall resistance was well established, the outside heattransfer coefficient, which is a combination of the condensation resistance and the coating resistance, can be computed from equation (3.3):

$$1/h_{0} = 1/U_{0} - A_{0}/h_{1}A_{1} - R_{w}$$
 (3.3)

During filrwise condensation on a smooth tube, the outside and inside resistances are of the same order of acculate. However, with dropwise condensation, it was anticipated that the cutside resistance would be much smaller than the inside resistance and the inside resistance would control the process. Therefore, small uncertainties in the determination of the inside resistance would produce relatively large variations in the computed value of the outside resistance. In order to minimize this effect, the inside coefficient was enhanced by utilizing a concentric spiral insert which ran the length of the tube. The increased turbulence produced by the insert resulted in a decrease in the internal thermal resistance. Utilizing a plain copper tube instrumented with six wall thermocouples and the spiral insert, the inside coefficient was determined using a Sieder-Tate-type correlation of the form shown in equation (3.4):

$$Nu = C_{Re}^{0} \cdot {}^{8}Pr^{1/3}(\mu/\mu_{L})^{0} \cdot {}^{14} + B \qquad (3.4)$$

Details regarding the determination of this coefficient are provided in Ref. 20 and Ref. 26.

#### C. FRYSICAL PROPERTIES TESTS

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Two standard testing procedures were employed to assess specific physical characteristics of each coating; ASTM specifications D 3359-78 <u>Measuring Adhesion by Tape Test</u> and D3363-74, <u>Film Hardness by Pencil Test</u>. Since both of these tests are destructive, it was not possible to subject specimens which were to be tested for endurance. Therefore, at least one specimen from each group of coatings was held back for destructive testing. The results derived from the testing one specimen per set were assumed to be representative of all samples in that set.

To provide statistically valid results which would ensure repeatability and reproducibility, these tests should be performed on a large number of samples. Due to the limited availability of many of the coatings, this discuss possible. These tests were performed to provide a relative indication of toughness and durability and to indicate the likelihood of the degradation of these characteristics upon exposure to steam for extended periods.

#### IV. RESULTS AND DISCUSSION

#### A. ENDUFANCE TEST

Euring this stage of the thesis, evaluation was limited to visual observation supported by periodic photographic sessions. With ideal dropwise condensation, the drops should appear spherical in shape, exhibit a large contact angle and grow to no more than two to three mm in diameter before departure. Departure from a vertical surface should be swift, straight down and the drop should retain its spherical shape. A flat appearance with irregularly-shaped edges during growth and the presence of a "tail" during departure are indications of "less-than-ideal" dropwise Although ideal dropwise condensation was the condensation. most desirable characteristic, durability and ease of applicaticn were also important considerations. Therefore, a of number coatings which exhibited somewhat less than ideal dropwise condensation were still considered for heattransfer measurements. A summary of all the coatings and the results of this stage of testing are provided in Table II.

# 1. No-Stik

Two identical sets, each consisting of four samples representing the four different substrates, were tested. Application and surface preparation were performed by the manufacturer. In order to verify previous observations, the second set was installed in the steam chamber approximately 2000 hours after the first set. Micrometers were used to determine a mean coating thickness of approximately 60 micrometers (0.0025 in). Due to the uneven texture of the



Figure 4.1 No-Stik and Nedox on NBr 0 hrs.



Figure 4.2 No-Stik Br Ti 790 hrs.



Figure 4.3 No-Stik Cu Br Ti 1500 hrs.





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Figure 4.4 No-Stik Cu Br 4000 hrs.

coating, an exact measurement of the coating thickness was not possible. The excellent quality of the dropwise condensation is readily apparent in Figures 4.1 through 4.4. The drops are nearly spherical, exhibit a relatively large contact angle, and grow to approximately 2.5 mm before departure. However, the rate at which the drops departed the surface was noticeably less than that of other specimens with thinner coatings. This indicated a lower heat flux caused by the thermal barrier imposed by the coating.

After more than 4000 hours of continuous exposure, this coating showed virtually no signs of physical degrada-The brass and copper samples were lightly speckled tion. with green which was attributed to oxidation of the copper. An examination of an "as delivered" sample with the SEM revealed small holes in the coating which exposed the Examination of an exposed specimen revealed substrate. that, although the coating was not affected, the substrate was being attacked at these void sites. However, this did not appear to affect the quality of the dropwise condensation, nor did the oxidation appear to undermine the coating in the immediate vicinity of the voids.

Eased upon the quality of the condensation and the durability of the coating, No-Stik was selected for heattransfer measurements.

# 2. <u>Nedcx</u>

A total of eight Nedox samples were tested, two sets of the four different substrates. Surface preparation and application were performed by the manufacturer. The coating thickness, supplied by the manufacturer, was approximately 5.0 micrometers (0.0002 in). This was confirmed by viewing a sectioned sample under the SEM.



Figure 4.5 Nedox Br Ti O hrs.



Figure 4.6 Nedox Br Ti Cu 264 hrs.



Figure 4.7 Nedox Ti Cu Blank 790 hrs.



Figure 4.8 Medcx Cu 2000 hrs.

As can be seen in Figure 4.5, the initial dropwise condensation was very good to excellent. This was true for all specimens. However, after forty-eight hours, all of the samples were noticeably darker in color, indicating a reaction of the substrate with the environment. At this point. the quality of the dropwise condensation was not noticeably After 240 hours, the condensation mode on the affected. upper-half of the naval brass specimen was filmwise. This specimen was removed for examination. After 1200 hcurs, only the copper specimen exhibited any degree of dropwise condensation. The changes in the quality of the condensation mode is readily evident from Figures 4.5 through 4.8. After 30CO hours, the remaining specimens were removed and the second set was installed. Although the exposure time varied, the pattern of performance was virtually the same; very good to excellent dropwise condensation occurred initially, with a gradual darkening of the substrate followed by a gradual degradation in dropwise performance.

Examination of an unexposed specimen under the SEM revealed a mottled surface covered by a thin, smooth, transparent film. The transparent film was presumed to be the Teflon coating while the mottled appearance was due to the chrome-nickel plating. Examination of an exposed specimen revealed that although the chrome-nickel plating was intact, the film had been peeled back or completely removed. Pieces cf the film still partially attached were visible with the The loss of the Teflon film accounted for the naked eye. loss of the hydrophcbic characteristics of the coating. However. the exact cause of this loss could not be determined.

Despite reservations about the durability of this coating, the excellent nature of the dropwise condensation made this coating a candidate for the heat-transfer experiments.

# 3. <u>Emralon 333</u>

A total of seven Emralon samples were tasted. Application and surface preparation were accompliable by the manufacturer of the coating. A set of four samples, one for each substrate, was initially loaded into the chamber while the second set of three samples, copper, titanium, and brass, was installed approximately 2000 hours later. The naval brass sample was withheld for destructive testing.

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Cn startup, all of the samples behaved predictably; condensation was mixed to filmwise. This behavior was anticipated because the fluorocarbon lubricants are bonded to the substrate with an organic resin binder which was not likely to possess a particularly low surface energy. Examination under the SEM revealed fluorocarbon particles imbedded in a resin matirx. However, with one exception, the mode of condensation gradually began to change after approximately 500 hours of exposure. Except for the copper sample, the condensation first turned from mixed to poor dropwise and then to good dropwise. After 1000 hours, the titarium, trass and naval trass samples were producing fair to excellent dropwise condensation (see Figures 4.9 through 4.12). An SEM examination of the exposed specimens revealed that the resin binder had been washed or ercded away. thereby exposing more of the imbedded fluorocarbons. This effectively lowered the critical surface tension of the condersing surface. The copper specimen never produced dropwise condensation. In fact, since faint copper colored areas could be seen with the naked eye, it was evident that the coating had been almost completely removed. These results were consistent for both sets of specimens.

Although the brass, titanium and naval brass specimens produced good to excellent dropwise condensation in excess of 4000 hours, Emralon 333 was not considered for



Figure 4.9 Emralon 333 Br Cu Ti 1500 hrs.



Figure 4.10 Emralon 333 Br Cu Ti 2000 hrs.



Figure 4.11 Earalch 333 Cu Ti 2700 hrs.



Figure 4.12 Emralon 333 Br Ti 3000 hrs.

heat transfer measurements because of the necessity of "ageing" the coating and the undesirable effect of contanination of the heat transfer apparatus due to the removal of the kinder.

# 4. Isonel 472

A total of eight Iscnel 472 samples were tested; four substrates, each with rough and smooth surface The initial condensation mode was characterized finishes. as fair to poor dropwise. The drops exhibited a relatively lcw contact angle, were irregular in shape, and grew to approximately 5.0 mm (0.2 in) before slowly departing the surface. After twenty-four hours of exposure, the condensation mode had changed to filmwise. This indicated a change in the critical surface tension of the condensing surface. This change was attributed to absorption of water by the coating. This result was the same for all samples. After 1000 hours, the samples were removed. No further observations were conducted.

#### 5. <u>Isonel 31-398</u>

A total of eight samples were tested. The results were virtually identical with those of the Isonel 472. No further observations were conducted.

The performance of the 472 and the 31-398 was not unexpected. These polyesters could be expected to present basically a hydrocarton surface to the condensing steam. Therefore, these compounds could be expected to have a relatively high critical surface tension which should produce relatively poor quality dropwise condensation. Other physical characteristics, such as ease of application and availability, made polyesters an attractive candidate. However, their inability to sustain dropwise condensation eliminated them from further consideration. No further tests were performed with these compounds.

# 6. <u>NEL C-6 Fluorcepoxy</u>

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Six C-6 samples were tested; substrates of copper, naval brass, and titanium were used with the two surflow finishes previously described. The brass samples were reserved for destructive testing. Because this compound is only available in limited quantities, application was by brush rather than by spraying or spinning. This resulted in an uneven surface texture estimated to be 5.0 to 10.0 micrometers thick. The estimate of the coating thickness was detarmined with micrometers and confirmed by viewing a sectioned portion of a specimen with the SEM.

All of the test specimens produced fair to good dropwise condensation. As can be seen in Figures 4.13 through 4.16, the drops are somewhat flat, irregular in shape, and grow to 4.0 mm (0.16 in) in diameter before slowly departing the surface. After 100 hours of exposure, the surfaces of the copper and naval brass specimens showed signs of oxidation. The oxidation grew progressively worse until after 1000 hours the surfaces were completely black. The oxidation of the surfaces of the reactive substrates indicated that the ccating was incapable of completely insulating the substrate from the environment. However, this did not seem to adversely affect the performance of the coating which remained virtually unchanged during the first 2000 hours of operation. After 2400 hours, all of the smooth samples exhibited some degree of separation of the coating from the substrate. After 3000 hours, the coating became unbonded from the smooth samples causing blisters and hare spots over 20% to 30% of the surface. This condition existed in excess of 4000 hours.

Although the cxidation of the copper and naval brass surfaces could be blamed for the deterioration of the adhesive qualities of the epoxy, this is not the case for

Figure 4.13 NRL C-6 Cu/S Cu/R 0 hrs.



Figure 4.14 NRL C-6 Ti/R Cu/S 790 hrs.



Figure 4.15 NRL C-6 Cu/S Cu/R 1500 hrs.



# Figure 4.16 NRL C-6 Cu/S Cu/R 4000 hrs.

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titanium. Bather, this condition seems to indicate an inadequate bond with the smooth surfaces. Althrugh the rough specimens fid not experience any delamination over the 4000 hours of testing, it is anticipated that the effect of the oxide layer will eventually have a deleterious upon the adherence of the coating.

Eased upon durability and good dropwise performance, this coating was selected for heat-transfer evaluation.

# 7. <u>NEL C-6' Flucroepoxy</u>

The specimen arrangement for the C-6' fluoroepoxy as well as the application and thickness of the coating ware identical to the C-6 epoxy. In addition, the endurance test results were virtually the same. The reactive surfaces experienced progressive oxidation, changing in color from a tright copper or brass to a uniform dull black. After roughly 1000 hours, the coatings on the smooth specimens showed signs of deterioration. Approximately 30% of the coating had been removed from the smooth titanium sample, while on the smooth copper and naval brass samples, although physically present, the coating had blistered and cracked allowing condensate to pass between the coating and the substrate. After 2400 hours, the rough specimens were still performing well. The departure size and shape of the drops did not differ markedly from the C-6. Therefore, this coating was not selected for heat-transfer measurements.

# 8. NRL BCE-7 Flucroepoxy

A total of six BCE-7 specimens were tested including both rough and smooth surfaces of the copper, naval brass and titarium substrates. Although this coating should have been highly hydrophotic because of the high concentration of -CF<sub>3</sub> groups, the dropwise condensation was judged to be only fair. Within two hours, the condensation mode had changed

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to filmwise. Closer examination revealed that over 90% of the coating had been removed from the substrate. The rapid deterioration of this coating was entirely unexpected. An explaration for the inadequate bonding of the coating to the substrate was not apparent. However, the performance was consistent for all substrates and surface finishes. No further tests were attempted with this coating.

### 9. NRL FlugIoacrylate

A total of six fluoroacrylate specimens were tested including both rough and smooth surfaces of the copper, titanium and naval brass substrates. As with the ECE-7 fluorcepcxy, this coating was virtually non-existent on any of the substrates within two hours. Since both coatings are they cannot polymers, be dissolved. therscsetting Therefore, this behavior appears to indicate an inadequate bond between the coating and the substrate. An adequate explanation for this behavior could not be discerned from the limited number of samples involved. No further testing of this particular coating was attempted.

# 1C. NRL Fluoroacrylic

As with the fluoroepoxies, six samples were tested. The three smooth samples failed almost immediately. Small patches of the ocating could be seen floating away with the condensate. However, all of the rough specimens produced good to excellent condensation in excess of 2500 hours with no signs of physical or hydrophobic degradation (see Figures 4.17 through 4.20). Although the drops are somewhat large, 3.8 mm (.15 in) in diameter before departure, they exhibit a larger contact angle and a more uniform shape than either of the epoxies. In addition, the oxidation rate of the reactive surfaces was noticeably less than either of the epoxy coated surfaces. This indicates that the acrylic was

Figure 4.17 NRL Acrylic Cu/S Ti/R 0 hrs.



Figure 4.18 NBL Acrylic Cu/R Cu/S Ti/R 700 hrs.



Figure 4.19 NRL Acrylic Cu/R 1000 hrs.



Figure 4.20 BRL Acrylic Cu/R Ti/R 2000 hrs.

better able to insulate the substrate from the environment. On the basis of performance and durability, the coacing was selected for heat-transfer measurements.

11. <u>FIFE</u>

A total of sixteen FTFF samples, two complete sets of four substrates and two surface finishes, were tested. Specizers from the first sat were tested for a total of 3000 hours before being replaced by the second set to verify previous observations.

Immediately after startup, all of the specimens produced excellent dropwise condensation. However, within four hours, the condensation mode on the brass and naval brass specimens had changed from dropwise to almost completely filmwise. The titanium samples also showed signs of failing. After forty-eight hours, only the copper specimens continued to promote dropwise condensation. After 240 hours of exposure, the non-performing specimens were removed for examination under the SEM.

The copper specimens were exposed in the test chamber for 3000 hours. The performance of these specimens during this period is shown in Figures 4.21 through 4.24 Throughout the test, the copper substrates showed signs of oxidation; their color gradually turned from a bright copper to a dull black. As can be seen in Figure 4.24, both copper specimers were performing poorly at the end of the test. At this point, the copper specimens were removed for examination and the second set was installed. None of the samples from the second set produced dropwise condensation after four hours of exposure.

Examination of various specimens under the SEM revealed different modes of failure. In the case of titarium, the PTFE was not adequately bonded to the substrate. After a short period of exposure, the coating flaked off in

Figure 4.21 Sputtered PTFE Cu/S Cu/R 0 hrs.



Figure 4.22 Sputtered FTFE Cu/S Cu/R 790 hrs.

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Figure 4.23 Sputtered PTFE Cu/S Cu/R 2000 hrs.



Figure 4.24 Sputtered PTFE Cu/S Cu/R 2700 hrs..

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small patches exposing the bare substrate. This was consistent for all of the titanium specimens. The failure mode of the brass and naval brass specimens was similar. Calv the copper specimens exhibited adequate adhesion throughout the duration of the test. However, SEM photographs revealed that the ultra-thin coating of PTFE was incapable of insulating the substrate from the environment. Numercus small, broker blisters caused by the oxidation of the substrate beneath the coating, covered much of the surface. This condition was revealed macroscopically by a gradual darkening of the surface and a degradation of the hydrophobic characteristics of the coating. The rapid failure of the second set can only be attributed to poor adhesion.

As pointed out in Ref 9, adhesion is dependent upon many factors some of which are not fully understood. Although some film-substrate combinations exhibit good adhesion, some of the more interesting combinations are incompatible in terms of adhesion. The results of this test indicate that the adhesion qualities of sputtered PTFE on the various substrates must be thoroughly investigated before utilizing this process to deposit thin films on condenser tubes. Eased upon these results, this coating technique was not pursued for heat transfer measurements.

## 12. Farylene-N

A total of eight samples, four different substrates with two coating thicknesses of 0.5 micrometers and 1.0 micrometer, were tested. The coatings were applied by the Lawrence Livermore National Laboratory on an "as available" basis. Due to scheduling difficulties, it was not possible to prepare the substrate surfaces prior to coating. Therefore, unlike the other specimens, none of the specimens to be coated with parylene-N underwent any surface preparation. All of the specimens were plated in the "as



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Figure 4.25 Parylene-N NBr 0 hrs.



Figure 4.26 Parylene-N NBr 800 hrs.

delivered" condition. For the copper, brass and titanium substrates the "as delivered" condition was similar to the "smooth" surface previously described. However, the naval brass was received with a surface similar to the "rough" surface of the prepared specimens.

Upon exposure to steam, seven of the specimens failed within twenty-four hours. Large, water-filled blisters covered as much as 90% of the substrate surfaces. The only specimen to perform satisfactorily was the onemicrometer-thick coating on the naval brass substrate. As can be seen in Figures 4.25 and 4.26, this coating produced dropwise condensation which could be characterized as fair to good. The drops are somewhat irregular but grow to only 3.5 mm (.14 in) in diameter before quickly departing the surface. After 1500 hours of exposure, this specimen showed no signs of physical or hydrophobic degradation.

Eased upon the limited number of parylene samples involved and observations of other coatings under similar conditions, failure can be attributed to two causes: surface finish and coating thickness. The fact that only the naval brass specimen with the 1.0 micrometer coating endured, indicates that a rough surface is necessary to insure proper This finding is consistent with observations of bonding. cther ccatings. In additicn, thicknesses of less than one micrometer appear to be incapable of withstanding the test environment. It is surmised that water vapor can penetrate the ultra-thin coatings in sufficient quantity to eventually undermine the bond between the coating and the substrate surface. This observation is supported by the failure of the 0.5 micrometer coating on the rough, naval brass substrate and the appearance of water filled blisters on all of the failed specimens. The bilsters indicated that, although the coating was continuous and firmly attached at the edges, condensate had somehow collected between the film and the substrate.

Eased upon the performance of the one good sample with consideration given to its unique application process, parylene-N was selected for heat-transfer measurements.

# 13. Farylene-D

The parylene-D specimens were received too late for the test results to be included in this thesis. However, sixteen parylene-D specimens, four substrates with both surface finishes and thicknesses of 0.5 and 1.0 micrometer are currently undergoing endurance testing during the writing of this thesis.

# 14. Ferco 6122

Since the application and substrate preparation for this coating were proprietary, only four specimens representing the different substrates were tested. Although the initial condensation mode was dropwise, it gradually turned to mixed condensation, part dropwise and part filmwise, within 100 hours. This remained the steady-state mode of condensation for over 2000 hours. The most likely explanation for the gradual change was the absorption and retention of water by the coating. This had the effect of gradually altering the critical surface tension. Although the coating did not promote dropwise condensation to a degree that would warrant further investigation, the coating did not suffer physical degradation by exposure to steam at atmospheric pressure in excess of 2000 hours.

15. <u>Silicone</u>

The silicone compound tested produced only fair dropwise condensation. The drops were relatively large, flat, and irregular in shape. In addition, the coating did not adhere to either the rough or smooth surfaces of any substrate. Blisters, filled with water, appeared within

twenty-four hours. These blisters continued to grow while undermining the coating and forcing it to separate from the substrate. Because the quality of the dropwise condensation was only fair and adherence appeared to be inadequate, no further testing was done.

#### E. FEYSICAL PROPERTIES TESTS

The results of the physical properties tests as well as the endurance test results are summarized in Table II.

## C. BEAT-TRANSPER RESULTS

A total of six ccatings were selected for heat-transfer measurements. Realizing that, for dropwise condensation, the determination of the outside heat-transfer coefficient is very sensitive to variations in the inside coefficient, extreme care was exercised in establishing a reliable, repeatable Sieder-Tate ccefficient for an internally enhanced smooth tube. A coefficient (C,) of 0.0638 with an intercept of (B) 26.9 was established for the Sieder-Tate equation through the analysis' of data taken from a plain copper tube instrumented with six thermocouples. Although not normally associated with the Sieder-Tate equation, the presence of an intercept was necessitated by the low L/D ratio of the test tubes. These values were used for detersining the inside heat-transfer coefficient for all subsequent data runs. A description of the data acquisition and reduction for determining these values is provided in Ref. 20 and Ref. 26. Figure 4.29 shows the variation of the Nusselt number as a function of the Sieder-Tate parameter for the instrumented tube. All of the data runs, a minimum cf twc fcr each coating, were made under vacuum conditions, approximately 0.012 MFa (1.6 psia). The mass concentration cf the ncr-condensing gases was maintained at ±0.5% (i.e.

TABLE II

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Endurance Test Results

Coating	Sukstrate/ Surface	Thick- ness m	Hard- ness	Adhesion	Dropwise Performance	Hours of Operation
No-Stik	Cu/U	60	t F	5 B	Excellent	>4000
No-Stik	Ti/U	60	4 F	5 B	Excellent	>4000
No-Stik	Br/U	60	4 F	5 B	Excellent	>4000
No-Stik	NEr/U	60	4 F	5B	Excellent	>4000
Nedcx	Cu/U	5	Å	5 B	Excellent	>2000
Nedox	Ti/U	5	£4	5 B	Excellent	<1000
Nedox	Br/U	J.	Ē4	5 B	Excellent	<4.0.0
Nedcx	NBI/U	5	ы	5 B	<sup>p</sup> xcellent	<300
Emralon 333	Cu/U	20	ß.	5 B	POOL	<500
Emralon 333	エュノロ	20	Ъ	5B	Excellent	>4000
Emralon 333	Br/U	20	ßı	5 B	Fair/Good	4000
Emralon 333	NET/U	20	œ.	5 B	Fair/Good	>4000
C-6 Flucioepoxy	Cu/S	1-10	2 H	5 B	Fair/Good	2000
C-6 Flucrospoxy	Cu/R	1-10	2 H	5 B	Fair/Gooð	>4000
C-6 Flucroepoxy	Ti/S	1-10	2 H	5 B	Fair/Gcod	2000
C-6 Flucroepoxy	Ti⁄R	1-10	2 H	5 B	Fair/Good	>4000
C-6 Flucroepoxy	NET/S	1-10	2 H	5 B	Fair/Gcod	2000
C-6 Fluctospcxy	NEL/R	1- 10	2 H	5 B	Fair/Good	>4000

TABLE II

# Endurance Test Results

Coating	Substrate/ Surface	Thick- ness m	Hard- ness	Adhesion	Dropwise Performance	Hours of Operation
C-6° Flucroepoxy	Cu/S	1-10	2 H	5 B	Fair/Good	1000
C-6 Flucroepoxy	Cu/R	1-10	2 H	5 B	Fair/Good	>2000
C-6 Flucroepoxy	Ti/S	1-10	2 H	5 B	Fair/Good	1000
C-6° Flucrospoxy	Ti/R	1-10	2 H	5 B	Fair/Good	>2000
C-6 Flucroepoxy	NEr/S	1-10	2 H	5 B	Fair/Gcod	1000
C-6 Flucroepoxy	NBL/R	1-10	2H	5.B	Fair/Good	>2000
BCE-7 Fluoroepoxy	cu/s	1-1 Q	НЕ	5 B	Failed	<20
Fluoroacrylate		1-10	НВ	4 B	Failed	<20
Flucioacrylic	Cu/S	1-10	لتر	3 B	Failed	<20
Flucroacrylic	Cu/R	1-10	Ç4	3 B	၆ငဝ႖ီ	>2500
Fluoroacrylic	Ti/S	1-10	<b>6</b> 44	3 B	Fail∈d	<20
Flucroacrylic	Ti/R	1-10	ŝų	3 B	Good	>250:)
<b>Fluoroacrylic</b>	N Br/S	1-10	Бч	3 B	Failed	<20
Flucroacrylic	NEL/R	1-10	ßı	3B	Gcođ	>2503
Pepco 6122	C u / U	5-10	НВ	5 B	Poor	>2000
Pepco 6122	Ti∕U	5-10	HB	5 B	POOL	>2000
Pecco 6122	NBIJU	5-10	НВ	5 B	Pcor	>2000

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TABLE II

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# Endurance Test Results

		-	•		-	
Coating	Substrate/ Surface	Thick- ness m	Hard- ness	Adheslon	Dropwise Performance	Hours of Operation
Isonel 472	All	5-10	Н	5 B	Failed	<24
Isonel 31-398	<b>All</b>	5-10	Н	5 B	Failed	<24
Sputtered PTFE	Cu/S	<b>.</b>	E.	4 B	Excellent	2000
Sputtered PTFE	Cu/R	4.	£4	4 B	Excellent	2000
Sputtered PTFE	Ti/S		ţzı	4 B	Excellent	<48
Sputtered PTFE	Ti/R	. 4	ítu	4 B	Excellent	<44 B
Sputtered PTFE	Br/S	<b>±</b> .	ĝ.	4 B	Excellent	<48
Sputtered PTFE	ET/R	4.	£4	4 B	Excellent	<48
Sputtered PTFF	NBI/S	<b>.</b>	βu	4 B	Excellent	<48
Sputtered PTFE	NEL/R	• 4	£4	4 B	Excellent	<48
Farylene-N	LIA	•2	Ø	4 B	Feiled	<20
Parylene-N	C u / A D	1.0	A	4 B	Failed	<100
Farylene-N	Ti/AD	1.0	83	<b>4</b> B	Failsd	<100
Parylene-N	BI/AD	1. 0	В	4 B	Failed	<100
Parylene-N	NBI/AC	1.0	B	4 B	Goođ	>1500
Silicone	All	D	4 B	5 B	POOL	<50

zero to within the accuracy of the temperature and pressure measurements) during all data runs.

The results of all the data runs with a least-squares curve fit are plotted in Figures 4.27 and 4.28. For comparison purposes, data representing the performance of a plain, smooth tube are also plotted. Because the program used for reducing and plotting the data was slightly different from that used in Ref. 20, a listing of the program (DAF3) and a sample of the reduced data are provided in Appendix A. Fhotographs of the tubes under actual test conditions are provided in Figures 4.30 through 4.33. The performance of each coating was as follows:

1. <u>No-Stik</u>

Even though this coating produced excellent dropwise condensation, the cutside heat-transfer coefficient was reduced by a factor of two-thirds when compared with a smooth tube undergoing filmwise condensation. This result is explained by the thickness of the coating which averaged 75 to 100 micrometers (0.003 to 0.004 in) thick. Although the coating was heavily doped with copper to improve conductivity, the thermal resistance imposed by the coating was far to be offset by the beneficial effects of dropwise condensation.

# 2. Nedcx

The Nedox coating improved the outside heat-transfer coefficient by 700% to 900%. This represented the best enhancement produced by any coating. However, questions as to the durability of this coating, which were raised during the endurance test, still remain. No degradation was evident during the heat-transfer tests.

# 3. FEIYlene-N

Two thicknesses of parylene-N were tested: 0.5 and 1.0 micrometer. The 0.5 micrometer coating enhanced the outside coefficient from 600% to 750%. As expected, due to the increased thickness, the 1.0 micrometer coating produced a lower enhancement of 500% to 600%. However, close inspection of the 0.5 micrometer coating after the heat-transfer tests, revealed small areas in which the coating was beginning to peel or crack. The one-micrometer coating showed no signs of deterioration. This result was consistent with the endurance test which indicates that 0.5 micrometers is too thin to produce reliable durability.

# 4. <u>C-6 Fluoroepcxy</u>

The NRL C-6 fluoroepoxy produced an improvement of 200% to 240%. Although considerably greater than a smooth tube undergoing filmwise condensation, these results are less than those achieved by fin tubes (See Ref. 20). This disappointing performance was probably a result of the coating thickness. Application by brush produces a coating which is slightly too thick. The thickness on the test tube was estimated to be 10.0 to 20.0 micrometers. Although very durable, an application technique which will consistently produce an ultra-thin, uniform coating must be utilized if this coating is to be exploited for enhancing heat transfer.

# 5. Flucroacrylic

Since the NRL fluoroacrylic is a thermcplastic polymer, it can be easily thinned prior to application. This makes a thin, uniform coating more easily obtainable when compared to the epoxies. In this case, the coating was estimated to be 5.0 to 10.0 micrometers. Therefore, the fluoroacrylic was capable of producing enhancements of 500% to 600%.



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Figure 4.30 Nedox.



Figure 4.31 NRL Fluoroacrylic.

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Figure 4.32 No-Stik.



Figure 4.33 Parylene-N.

# V. CONCLUSIONS AND RECOMMENDATIONS

### A. CONCIUSIONS

1. During the cordensation of steam on horizontal tubes, the cutside heat-transfer coefficient can be enhanced five to eight times through the use of organic polymer coatings. A 500% increase can be achieved with coatings of demonstrated durability, (i. e. coatings which showed no degradation over a minimum period of 2000 hours), whereas the higher rates can be achieved only by coatings of questionable durability.

2. Because of their hydrophobic characteristics, ease of application and the ability to be applied in ultra-thin layers, the fluoroepoxies and fluoroacrylics represent the best alternative for the application of a low energy, fluorocarton surface.

3. Both the Nedox coating and the parylenes offer possible alternatives to the epoxies and acrylics once their durability is firmly established.

4. A rough surface is essential for the proper bonding cf the polymer coatings tested.

5. The ultra-thin polymer coatings tested were incafable of completely insulating reactive substrates from the environment.

6. The outside heat-transfer coefficient is highly sensitive to uncertainties in the inside coefficient. Therefore, the inside coefficient must be clearly established before attempting to infer the outside coefficient from overall measurements.

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#### E. BECCEMENDATIONS

1. Continue the evaluation of the fluoroepoxies and fluoroacrylics. New compounds have been developed and these previously tested can be modified to enhance durability, while maintaining most of their hydrophobic characteristics.

2. Devise a method for applying a controlled, uniform, repeatable thickness for the fluoroacrylics and fluorcepoxies.

3. Continue to evaluate the parylenes for durability. When compared to the other parylenes, parylene-C has relatively low water vapor and oxygen transmission rates. Therefore, parylene-C should be included in the evaluation.

4. Attempt to reduce the thickness of the No-Stik coating.

5. Determine and eliminate the cause for the deterioration of the NEDOX coating.

6. Apply selected coatings to tubes made of materials with low thermal conductivities, such as stainless steel or titanium, to determine the effect of the constriction resistence upon thermal performance.

7. In order to prevent exidation and eventual coating failure, use non-reactive substrates. If this is not possible, the substrate should be plated with a non-reactive sub-layer.

# APPENDIX A COMPUTER PROGRAM LISTING

The following pages contain a listing of the computer program used for data acquisition, data reduction and plotting.

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```
1)011 - DLE HAME: 0663
10101 REVISED: December 19, 1983
10291
          CCH /CC/ C(7)
DIM Emf(10)
DATA 0.19986091.25727.94369.-767345.9295.78025595.91
DATA -9247486589.6.97688E+11.-2.66192E+13.3.94078E+14
1030
1040
 1050
1060
          READ C(+)
1070
          Di=.0127
Do=.01905
                                  ! Inside diameter of test tube
! Dutside diameter of test tube
! Dutside diameter of the outlet end
1020
1090
1100
           Dr . 015875
          Dspp=1524 ! Inside drameter of stainless steel test section
Ax*PI+Dsp 2/4-PI+Do+L
L=.130175 ! Condensing length
1110
1129
          Ax*P1+Desp 2/4-P1+Dest
L=.130175 1 Condensing length
L1=.050325 1 Enlet end "fin length"
L2=.034925 1 Outlet end "fin length"
Kcu=365 1 Thermal conductivity of Copper
Rm=Do+LOG(Do/Di)/(2*Kcu) ! Wall resistance based on outside area
1140
1150
1150
1130
           PRIVIER
                         ISI
1190
           CLEAR 709
1200
1210
1220
           BEFP
          PRINT USING "4X.""SELECT OPTION:"""
PRINT USING "6X.""1 Taking data or re-processing previous data"""
PRINT USING "6X.""2 Plotting data"""
1230
          INPUT IOP
PRINTER IS 701
1240
          IF IOP=2 THEN 4620
BEEP
1250
1250
1270
1280
1290
           DEEF
THPUT "ENTER HONTH, DATE AND TIME (MM:DD:HH:MM:SS)",Dates
OUTPUT 709:"TD":Dates
OUTPUT 709:"TD"
1300
          ENTER 709:Dates
 1310
1320
                                         Month. Jate and time :":DateS
           PRINT
  341)
           PRINT USING "10X,""NOTE: Program name : DRP3"""
1350
           REFE
           HIPUT "ENTER DISK NUMBER".Dn
PRINT USING "15X.""Disk number = "".DD":Dn
1360
 1370
          INPUT "ENTER INPUT MODE (1=3054A.2=FILE)".Im
IF Imal THEN
REEP
 1380
 1990
1400
 iaiŋ
          PEEP
INFUT "GIVE A NAME FOR THE RAW DATA FILE".D_file$
PRINT USING "16%.""File name : "".14A":D_file$
"PEATE EDAT D_file$.15
AGSIGN #File TO D_file$
1420
1430
1 4 4 0
1450
1460
           Ifa=1
1470
           Innel
          IUT-0
OUTPUT %File:Ifg.Inn
IF Ifg-0 THEN OUTPUT @File:Iwt
ELCE
BEEP
1 4 8 0
 14:30
1500
1510
1520
1530
           INPUT "GIVE THE NAME OF THE EXISTING DATA FILE",D_file$
PRINT USING "16X.""This analysis was performed for data in file "".10A":D_
1540
File5
1550
           BEFF
:560
           INPUT "ENTER THE NUMBER OF RUNS STORED" .Nrun
          ASSIGN #File ID D_files
ENIER #File ID D_files
ENIER #File:Ifg.Inn
IF Ifg=0 THEN ENTER #File:Iwt
END IF
BEEP
1570
1590
1500
1610
```

\_\_INPUT "CHIER GETICH (ted) T.PateFILE,3=HVE)".[tm bere 15.0 14.0 INPUT "ENTER UPTION FOR END of INTERNET THE ADDED THEM 14.0 INPUT "ENTER UPTION FOR END of INTERT." This analysis uses QCT readings""" 1650 IF Itm=1 THEN PRIMI USING "16X." This analysis uses T-PILE readings""" 1670 IF Itm=3 THEN PRIMI USING "16X." This analysis uses average of QCT and T-P ILE readings"" 1680 IF Ife=1 THEN PRIMI USING "16X." This analysis includes end-fin effect""" 1690 IF Ife=1 THEN PRIMI USING "16X." This analysis includes end-fin effect""" 1690 IF Ife=1 THEN PRIMI USING "16X." This analysis includes end-fin effect""" 1690 IF Ife=0 THEN PRIMI USING "16X." This analysis neglects end-fin effect""" 1700 IF Ife=0 THEN PRIMI USING "16X." This analysis neglects end-fin effect""" 1710 IF Ife=0 AND Inn=0 THEN CI=.0294 1710 IF Ife=0 AND Inn=0 THEN CI=.070 1720 IF Ife=1 AND Inn=1 THEN CI=.070 1720 IF Ife=1 AND Inn=3 THEN CI=.042 1720 IF Ife=0 AND Inn=3 THEN CI=.042 1750 FRIME USING "16X." Sider-Tate coefficient = "".Z.4D":Ci 1770 BEEP 1760 BEED INPUT "GIVE A MAME FOR PLOT DATA FILE".P\_files CREATE BDAT P\_files.5 ASSIGN @Filep TO P\_files 1780 1790 1300 1810 REFE INPUT "ENTER OUTPUT VERSION (1=SHORT,2=LONG)", Iov 1820 IF Iovel THEN PRINT IF Infel THEN PRINT USING "10X,""Data Yw 1830 1340 1850 1360 F 1370 Uo Ho ЯD Vv Nr 1880 PRINT USING "10X,"" # (m/s) (W/m 2-K)(W/m 2-K) (W/m 2) (m/s)""" ELSE PRINT USING "10X.""Data Vw PRINT USING "10X."" # (m/s) 1890 9...... Uo (W/m 2-K) Ho Ωp 1900 1910 (m/s)"" (W/m'2-K) (W/m 2) 1920 END IF 1930 END IF 1940 So\_on= 60\_0n=1 1950 Repeat:! 1360 0k 3+1 J=J+1 IF Im=1 THEN BEEP 1970 1990 2000 INPUT "LIKE TO CHECK NG CONCENTRATION (1=Y.O=N)?".Ng 2010 BEEP BEEP INPUT "ENTER FLOWMETER READING".Fm OUTPUT 709:"AR AF50 AL63 VR5" OUTPUT 709:"AS SA" ENTER 709:Etp DUTPUT 709:"AS SA" 2030 2040 2050 2060 Vtran=0 FOR I=1 TO 50 ENTER 703:Vt Vtran=Vtran+Vt 2070 180 2090 2100 2110 2120 2130 NEXT I Vtran=Vtran/50 OUIPUT 709:"AS SA" 2140 BEEP 2150 INPUT "CONNECT VOLTAGE LINE", OK 2160 ENTER 709:Bvol BEEP INPUT "DISCONNECT VOLTAGE LINE".Ok 2170 2190 OUTPUT 709: "AS SA"





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MICROCOPY RESOLUTION TEST CHART NATIONAL BUREA OF STANDARDS-1963-A

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2200 2210 2220 2230 2240 2250 ENTER 709:84mp DUTPUT 709:"AR AF20 AL24 VR1" FOR I-0 10 4 DUTPUT 709:"AS SA" Se=0 FOR K=0 TO 10 ENTER 709:E 2250 FUK K-0 10 10 2260 ENTER 709;E 2270 Se-Se+E 2280 NEXT K 2290 Emf(I)=ABS(Se/10) 2300 NEXT I 2310 OUTPUT 709;"AS SA" 2320 OUTPUT 713;"TIR2E" 2330 HAIT 2 2340 ENTER 713;T11 2350! OUTPUT 713;"T2R2E" 2360! HAIT 2 2370! ENTER 713;T2 2380 BEEP 2390 INPUT "ENTER T2", T2 2400 T2-9.0378E-2+T2=1.0034511 2410 OUTPUT 713;"T1R2E" 2420 HAIT 2 2430 ENTER 713;T12 2440 T1=(T11+T12)=.5 2450! OUTPUT 713;"T3R2E" 2460 IF Ng=0 THEN 2590 2470 BEEP 2480 INPUT "ENTER MANOMETER REA 2490 BEEP 2250 2260 2270 2280 2290 2290 2310 2310 DEEP INT TO ACCEPT THIS RUN (1=Y-DEFAULT, 0=N)", DK 3 INPUT "DK TO ACCEPT THIS RUN (1=Y-DEFAULT, 0=N)", DK 3 IF DK 3=0 THEN J=J-1 2480 INPUT "ENTER MANOMETER READINGS (HL.HR 2480 INPUT "OK TO ACCEPT THIS RUN (t=y=DEFA2510 INPUT "OK TO ACCEPT THIS RUN (t=y=DEFA2510 IF Ok 3=0 THEN 2520 J=J=1 2530 GDTO 1960 2540 END IF 2550 Phg=H1+Hr 2560 Phg=H1+Hr 2560 Phg=H1+Hr 2570 ELSE 2580 ENTER  $\Im F_1 le:Bvol.Bamp,Vtran.Etp.Emf(0)$ 2.Phg.Pwater 2530 IF J=1 OR J=10 OR J=20 OR J=Nrun THEN 2600 Ng=1 2510 ELSE 2620 Ng=0 2500 END IF 2640 END IF 2640 END IF 2650 Isteam=FNIVsv(Emf(0)) ! COMPUTE STEAM INPUT "ENTER MANOMETER READINGS (HL.HR,HRW)".H1,Hr,Hrw ENTER OFile:Bvol.Bamp, Vtran.Etp.Emf(0), Emf(1), Emf(2), Emf(3), Emf(4), Fm.T1, T END IF Tsteam=FNTvsv(Emf(0)) ! COMPUTE STEAM TEMPERATURE Troom=FNTvsv(Emf(3)) IF Iut=1 THEN Twm=0. FOR I=11 TD 5 Tw(I)=FNTvsv(Emf(I+5)) Twm=Tym+Tw(I) NEXT I Twm=Tym+Tw(I) NEXT I Tcon=FNTvsv(Emf(4)) Psat=FNPvstTsteam) Rohg=13529=122\*(Troom=25.95)/50 Rowater=FNRhow(Troom) 26500 26500 26500 26500 2700 27700 27730 27730 27750 2770 2780 Rowater=FNRhow(Troom)

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Ptest=(Phg=Rohg=Puater=Rowater)=9.799/100 Pmm=Ftest/133.322 Pkm=Ptest=1.E=3 Pks=Psat=1.E=3 Pkt=FNPvsv(Vtran)=1.E=3 Tsat=FNTvsp(Ptest) Vst=FNVvst(Tsteam) Ppng=(Ptest=Psat)/Ptest Post=1=Ppng Mfng=1/(1+18.015/28.97=Psat/(Ptest=Psat)) Vfng=Mfng/(1.608=.608=Mfng) Mfng=Mfng=100 Vfng=Vfng=100 BEEP IF\_Iov=2 THFM 2790 Ptest\*(Phg\*Rohg-Puater\*Rowater)\*9.799/1000 2866 2820 2830 2840 2850 2860 2870 2880 2890 2910 2920 2930 2940 2950 IF Iov=2 THEN PRINT PRINT USING "10X,""Data set number - "".DD":J END IF IF IOV-2 AND Ng=1 THEN PRINT USING "10X."" 2960 2350 END IF 2370 IF Iov-2 AND Ng=1 THEN 2980 PRINT USING "10X."" G %""" 2390 PRINT USING "10X."" (mm) Ρ Psat Ν Ptran Ineas Tsat (kPa) (kPa) (kPa) (C) (0) Molal Mass 3000 PRINT USING "10X.5(30.DD.2X).2(30.DD.2X).2(M3D.D.2X)":Pmm.Pkm.Pks.Pkt.Tste am.Tsat.Vfng,Mfng 3010 PRINT END IF IF Mfng>.5 THEN 3020 3830 BEEP 3040 3050 PRINT IF Im=1 AND Ng=1 THEN BEEP PRINT 3060 3070 3080 3090 PRINT USING "10X.""Energize the vacuum system """ 3100 BEEP 3110 3120 3130 INPUT "OK TO ACCEPT THIS RUN (1=Y.0=N)?", Ok IF OF =0 THEN DISP "NOTE: THIS DATA SET WILL BE DISCARDED!! " MAIT 5 GOID 1980 3140 3150 3160 END IF 3170 3190 END IF 3190 3200 3210 3220 3230 3230 3240 3250 IF Im=1 THEN IF Fm<10 DR Fm>100 THEN Ifm=0 BEEP INPUT "INCORRECT FM (1~ACCEPT.0-DELETE)".Ifm IF Ifm=0 THEN 1960 END IF 3250 3260 END 3270 IF Jfg=1 3280 OUTPUT # 12,Phg.Pwater 7290 END IF TF Ifg= IF Jfg=1 DR lwt=0 THEN DUTPUT @File:Bvol.Bamp,Vtran.Etp.Emf(0),Emf(1),Emf(2),Emf(3).Emf(4),Fm.T1. 3290 END IF 3300 IF Ifg=0 AND Iwt=1 THEN OUTPUT @File:Bvol,Bamp,Ytran.Etp.Emf(+),Fm.T1.T2,P hg,Pwater 3310 END IF 3320 IF Ifg=0 AND Iwt=1 THEN OUTPUT @File1:Tw(+) 3320 IF Ifg=0 AND Iwt=1 THEN OUTPUT @File1:Tw(+) T1=FNTvsv(Emf(2)) 3340

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Grad=FNGrad((11+12)+.5) Io=1:+ABS(Etr)/(10=Grad)+1.E+6 Er1=ABS(1:-T1) PRINT USING """T1 (GCT) ..." PRINT USING """T1 (IC) ..." IF Er1>.5 THEN BEEP PRINT "GCT OND TO STORTE .... 3350 Grad=FNGrad((f1+T2)+.E) 3050 3370 3380 - "".DD.3D";11 - "".DD.3D";11 3390 3400 3410 3420 PRINT "OCT AND TO DIFFER BY MURE THAN 0.5 C" 3430 3440 BEEP BEEP INFUT "OK TO GO AHEAD (1=Y.0=N)?".0k1 END IF PRINT USING """DT (0CT) = "".Z.3D":F2-11 PRINT USING """DT (T-PILE) = "".Z.3D":F2-11 IF 0k10 AND Er12.5 THEN 4640 Er2=ABS((T2-T1)-(T0-T1))/(T2-T1) IF Er22.05 THEN 3450 3460 0470 3480 3490 3500 3500 3510 3520 3530 3540 PEEP PRINT "OCT AND T-PILE DIFFER BY MORE THAN 5%" HELP INPUT "OK TO GO AHEAD (1-Y.O-N)?".Ok2 IF Ok2=0 AND Er2>.05 THEN 4640 END IF 3550 3560 3570 PRINTER IS 701 IF Itm=1 THEN I11=11 3580 3590 3600 3610 3620 3630 T20-T2 END IF IF Itm=2 THEN T11=T1 3640 3640 111=11 3650 120=To 3660 END IF 3670 IF Itm=3 THEN 3680 111=(T1+T1)=.5 3690 120=(T2+T0)=.5 3700 END IF 3710 END IF END 1+ Tavg=(111+120)+.5 Cpw=FNCpw(Tavg) Rhow=FNRhow(Tavg) Md=5.00049E-3+6.9861937E-3+Fn Md=Md=(1.0365-1.96644E-3+Tavg+5.252E-5+Tavg-2)/.995434 Mf=Md/Rhow 3710 3720 3730 3740 3750 3770 Vw=MF/(PI+D1 2/4) 3730 IF Inn=0 THEN T20=T20-(.0138+.001\*Vw'2) 3730 IF Inn=1 THEN T20=T20-(.0138+.001\*Vw'2) 3800 IF Inn=1 THEN T20=T20-.0018\*Vw'2 3810 0=Md\*Cpu\*(T20=T11) 3820 0p=0/(PI=Do\*L) 3820 0x=5N#CT20= 3830 Kw=FNKw(Tavg) Muw=FNMuw(Tavg) Re:#Rhow\*Vw®Di/Muw Prw=FNPrw(Tavg) Fel=0. 3840 3950 3860 2870 3880 Fe2=0. 3890 Cf=1. 3880 3900 Hi=Kw=Ci/Di=Rei .8+Prw .3333=Cf 3910 Dt=Q/(P[=Di=(L+L1=Fe1=L2=Fe2)=Hi) 3920 Cfc=(Muw/FIMmw(Tavg=Dt)) .14 3930 IF ABS((Cfc-Cf)/Cfc)>.01 THEN 3940 Cf=(Cf+Cfc)=.5

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6010 3900 3950 GD10 3900 3960 END IF 1970 IF 150 0 1965 3017 1960 3980 P1+P1+(D1+D0) 3990 A1+(D0-D1)+P1+(D1+D0)+.5 4000 M1+(H1+P1/(Kcu+A1)) .5 4010 P2+P1+(D1+D1) 4020 D1+P1+(D1+D1) A2+(Dr-D1)+PI+(D1+Dr)+.5 M2+(H1+P2/(Kcu+A2)) .5 Fe1=FNTanh(M1+L1)/(M1+L1) Fe2=FNTanh(M2+L2)/(M2+L2) Fe2+FN1ank(M2\*L2)/(M2\*L2) Lmtd\*(f2o-f1)/L06((fsteam-f1))/(fsteam-f2o)) Horl/(Ln+d+P1\*Do\*L) Horl/(1/Uo-Do\*L/(D1\*(L+L1\*Fe1+L2\*Fe2)\*H1)-Rm) Dtc\*Q/(P1\*D1\*(L+L1\*Fe1+L2\*Fe2)\*H1) IF ABS((Dtc-D1)/Dtc)).01 THEN 3900 Hfg=FNHfg(Tsteam) OUTPUT #Filep:Qp.Ho O1\*S00 .:029 01-500 Gloss=Ol/(100-25)=(Tsteam=Troom) Hfc=FNHf(Tcon) Hf=FNHf(Tsteam) Mdv=0 Haves Bp=(Bvol=100) 2/5.76 Mdvc=((Bp-Qloss)-Mdv+(Hf-Hfc))/Hfg IF ABS((Mdv-Mdvc)/Mdvc)>.01 THEN Mdv=(Mdv+Mdvc)\*.5 GOTO 4190 GUID 4190 END IF Mdv=(Mdv+Mdvc)=.5 Vg=FNVvst(Tsteam) Vv=Mdv=Vg/Ax IF lov=2 THEN PRINT USING "10X."" I (Inlet) Delta-[""" PRINT USING "10X."" OCT IC OCT T-PILE""" PRINT USING "10X."" OCT IC OCT T-PILE""" PRINT USING "10X."" VW Rei Hi Uo Vv"" 4240 4250 4270 4280 Но a PRINT USING "10X.Z.DD. 1X.5(MZ.3DE. 1X).MZ.DD": VW.Rei.Hi.Uo.Ho.Qp.Vv 4330 END IF 4340 IF Iov=1 THEN 4350 IF Inf=1 THEN 4360 PRINT USING "11X.DD.2X.Z.DD.2X.2(5D.p.2X),Z.3DE.1X,Z.DD.2(1X.3D.DD)";J.Yw. Uo.Ho.Qp.Vv.F.Nr 4370 ELSE 4380 PRINT USING "11X.DD.2X.Z.DD.2X.2(MD.4DE.2X).Z.3DE.3X.Z.DD":J.Vw.Uo.Ho.Qp.V 4<u>3</u>30 END IF IF Im=1 THEN BEEP INPUT "WILL THERE BE ANOTHER PUN (1+Y.0+N)?", Go\_on Nrun=J Nrun-J IF Go\_on=1 THEN Repeat ELSE IF JKNrun THEN Repeat END IF IF Im=1 THEN BEEP PRINT PRINT USING "10X.""NOTE: "".ZZ."" data runs were stored in file "".10A":J. 

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4500 EHD IF 4540 BEEP 4550 PRINT 4560 PRINT USING "YOX.""NOTE: "".ZZ."" X-Y pairs were stored in plot data file "".10A":J.P\_file5 4570 ASSIGN #File FO \* 4580 ASSIGN #File FO \* 4580 ASSIGN #File FO \* THPUT "LIKE TO PLOT DATA (1-Y.O-N)?".OK IF OL+1 THEN 4600 4610 4620 4630 CALL Plot END IF END DEF FNPvst(Tsteam) 4640 4650 4550 DIM K(8) DATA -7.591234564.-26.08023636.-168.1706546.64.23285504.-118.3646225 DATA 4.16711732.20.9750676.1.69.5 4679 4620 4690 READ K(+) 4700 T=(Tsteam+273.15)/647.3 4710 4720 4730 Sum=0 FOR N=0 TO 4 Sum=Sum+K(N)+(1-T) (N+1) 4740 4750 NEXT N Br=Sum/(T+(T+K(5)+(1-T)+K(6)+(1-T)^2))-(1-T)/(K(7)+(1-T)^2+K(8)) Pr=EXP(Br) P=22120000+Pr 4750 4770 4780 RETURN P 4790 FNEND DEF ENHFg(T) Hfg=2477200-2450\*(T-10) RETURN Hfg 4800 4810 4820 4830 FILEND DEF FNMuw(T) A=247.3/([+133.15) Mu=2.4E-5+10 A 4840 4860 4870 RETURN Mu FILEND DEF FNVvst(Tt) P+FNPvst(Tt) T+Tt+273.15 X+1500/T F1=1/(1+1+1.E-4) F2=(1-EXP(-X)) 2.5=EXP(X)/X .5 B+.0015=F1-.000342=F2-.0004882=X K=2=P/(461.52=T) V=(1+(1+2=B+K)^.5)/K RETURN V RETURN Mu 4680 4300 4310 4920 4940 4950 4960 4980 4990 FNEND DEF FNCpw(T) Cpw=4.21120858-T+(2.26826E-3-T+(4.42361E-5+2.71428E-7+T)) RETURN Cpw=1000 5000 5020 5030 FNEND DEF FNRhow(T) Ro=999.52946+T+(.01269-T+(5.482513E-3-T+1.234147E-5)) RETURN Ro 5040 5050 5060 5070 FNEND 5090 DEF FNPrw(T) 5090! Prw='0 (1.09976605-T+(1.3749326E-2-T+(3.968875E-5-3.45026E-7+T))) 5100 Prw+FNCpw(T)+FNMuw(T)/FNKw(T) 5110 RETURN Pris

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SALD SHEND 5120 ETERNO 2120 DEF ETERNO 21400 Kw=.5525894+F\*(2.29F3546E-3-T\*(1.509756E-5-4.0581652E-8\*T)) 5150 X\*(T+273.15)/273.15 5150 Kw=-.92247+X\*(2.8395-X\*(1.8007-X\*(.52577-.07344\*X))) 5170 RETURN Kw 5190 ETERN FNEND DEF FNTanh(X) P=EXP(X) Q=EXP(-X) Tanh=(P-Q)/(P+Q) RETURN Tanh RETURN Tanh FNEND DEF FNTvsv(V) CDM /Cc/ C(7) T-C(0) FOR T=1 TD 7 T=T+C(T)+V'T NEXT T T=T+4.73386E-3+T+(7.592834E-3-T+8.077927E-5) RETURN T RETURN T ENEND DEF ENHF(T) HF=T=(4,203849=T=(5,88132E-4-T+4,55160317E-6)) RETURN HF+1000 FNEND DEF FNGrad(T) Grad=37.9853+.104388+T RETURN Grad 5390 5400 FNEND DEF FNTvsp(P) Tu+110 5410 5420 - 430 5440 5450 TI=10 TI=10 Ta=(Tu+TI)\*.5 Pc=FNPvst(Ta) IF ABS((P-Pc)/P)>.001 THEN IF Pc(P THEN TI=Ta IF Pc)P THEN Tu=Ta GDTO 5450 5460 5470 END IF END IF RETURN Ta FMEND DEF FNPvsv(V) P=8133.5133+2.236051E+4\*V RETURN P FNEND SUB Plot DIM C(9) PRINTER IS 705 BEEP BEEP INPUT "OK TO SELECT DEFAULT VALUES FOR CAGE ".OFd IF Ord-1 THEN BEEP INPUT "ENTER PRESSURE CONDITION (1-9.2-A)". Iva END IF 5670 5630 5630 5700 INPUT "SELECT OPTION (1=Ho.2=Dt)". The IF Ord=1 THEN IF Iva=1 AND Int=1 THEN 5710 Xmin=100000 5720 Kmax=600000

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5730 5740 5750 Ymax=120000 Xstep=100000 Ystep=20000 END\_IF 5760 5770 5790 5790 5800 TF Iva=2 AND Int=1 THEN Xmin=400000 Xmax=1.6E+6 5810 5820 Ymin=0 Ymax=120000 5830 Xstep=400000 Astep=400000 END IF IF Iva=1 AND Iht=2 THEN 5850 5860 5870 Xmin=0 5880 5890 5900 Xmax = 30 Ymin=100000 Ymax=600000 Xstep=10 5910 5920 5930 5940 Ystep=100000 END IF IF Iva=2 AND Int=2 THEM 5950 Xnin=D 5960 5970 Xmax=30 Ymin=400000 Ymax=1.6E+6 5980 5990 Xstep=10 E000 5010 Ystep=400000 END IF END IF 6020 IF OK -O THEN BEEP 6030 6040 5050 INPUT "ENTER MINIMUM AND MAXIMUM X-VALUES".Xmin.Xmax 6060 BEEP INPUT "ENTER MINIMUM AND MAXIMUM Y-VALUES", Ymin.Ymax 5070 8EEP 6080 6190 THPUT "ENTER STEP SIZE FOR X-AXIS". Xstep 6100 BEEP THPUT "ENTER STEP SIZE FOR Y-AXIS" . Ystep END IF BEEP 6120 SELP PRINT "IN:SP1:IP 2390,1800.8300.6800:" PPINT "SC 0.100.0.100:IL 2.0:" Sfg=100/(Xmax-Xmin) Sfg=100/(Ymax-Ymin) BEEP Icon0 6130 5140 6150 6160 6170 BEÉP Icg=0 IMPUT "LIKE TO BY-PASS CAGE (1-Y.0-N-DEFAULT)?".Icg IF Icg=1 THEN 6650 PRINT "PU 0.0 PD" FOR Xa-Xmin TO Xmax STEP Xstep X+(Xa-Xmin)\*Sfx PRINT "PA":X.".0: XT:" NEXT Xa PRINT "PA 0.0.0 PD" FOR Ya-Ymin TO Ymax STEP Ystep Y=(fa-Ymin)\*Sfy PRINT "PA 0.":Y."YT" HEXT Ya 6290 6300 6310 6320 6330 NEXT Ya PRINT "PA 0,100 TL 0 2"

Ymin=N

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FDP YasYmin TO Ymax CIEP kstat K:(YarYnin)\*GFx PPINT "PA'(X.".100): X'" MEXT Ya PRINT "PA r00.100 PU PA 700.0 PD" FOR Ya\*Ymin TO Ymax SIEP Ystep Y\*(YarYmin)\*SFy PRINT "PD PA 100.".Y."YT" NEXT Ya PRINT "PA 00.100 PU" PRINT "PA 0..2 SP 1.5.2" FOR Ya\*Gmin TO Ymax SIEP Xstep X\*(YarXmin)\*SFy PRINT "PA":X.".0:" IF Int=1 THEN PRINT "CP -4.-1:LB":Xa:"" IF Int=1 THEN PRINT "CP -4.-1:LB":Xa:"" IF Int=1 THEN PRINT "CP -4.-1:LB":Xa:"" IF Int=1 THEN PRINT "CP -2.-1:LB":Xa:"" PRINT "PA 0.":Y."" P ERAD FOR Kaskmin TO Ymax STEP Astab 5359 6350 6370 6380 6390 6400 6410 6420 6430 6440 645U 6460 6470 6480 6490 6500 6510 6570 6520 6530 6540 6550 6550 6570 5580 5580 5580 INPUT "ENTER X-LABEL", Xlabel\$ BEEP BEEP INFUT "ENTER Y-LABEL".Ylabel\$ PRINT "SR 1.5.2:PU PA 50.-10 CP":-LEN(Xlabel\$)/2:"0:LB":Xlabel\$:"" PRINT "PA -18.50 CP 0.":-LEN(Ylabel\$)/2=5/6:"DI 0.1:LB":Ylabel\$:"" PRINT "CP 0.0 DI" 5600 5610 6620 6630 5640 Repeat: ! 4650 5650 BEEP INPUT "LIKE TO PLOT DATA FROM A FILE (1=Y,0=H)?".0k 6670 5680 DEPTO DEPTO IF OK+1 THEN BEEP INPUT "HANT TO CHANGE PEN (1+Y.O-N)".OKP 6690 6700 INFUL "WANT TO IF OLD=1 THEN BEEP 6710 6720 6730 6740 PRINT "SPO" BEEP INPUT "CHANGE PEN AND HIT CONTINE".Okc PRINT "SP1" 5750 5750 £770 END IF 5780 INPUT "ENTER THE NAME OF THE DATA FILE".D\_Files 6790 ASSIGN @File TO D\_file® 6800 6810 Sx=0 Sy=0 6820 6330 5:2-0 5840 5850 Sxy 0 BEEP INPUT "ENTER THE BEGINNING RUN NUMBER" .Md 6860 5870 5880 BEEF INPUT "ENTER THE NUMBER OF X-Y PAIRS STORED" .NPAIRS 5890 5200 BEFF BEEP INPUT "SELECT A SYMBOL FOR THE PLOITER (1++,2++,3+c,4+o,5+')",Sym PRINT "PU DI" IF Sym+1 THEN PRINT "SN+" IF Sym+2 THEN PRINT "SN+" IF Sym+3 THEN PRINT "SNc" 691n 6920 6930 6.940

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6950 IF Sym=4 IHEN PRINT "DN:" 6360 IF Sym=5 THEN PPINT "SN " 6970 IF Md>Npairs THEN 6980 FOR I=1 TO (Md-1) 6030 ENTER @File:Xa.Ya 7010 MEXT I 7010 END IF 7020 FOR I=1 IO Npairs 7030 ENTER @File:Xa.Ya ENTER SFile:Xa.Ya 7959 Yt=Xa Xa•Xa/Ya Ya=Yt Yc=LUG(Ya) Xc=LDG(14) Xc=LDG(Xa) Sx=Sx+Xc Sy=Sy+Yc Sx2=Sx2+Xc<sup>+</sup>2 7090 7100 7110 7120 7130 7140 7150 Sxy=Sxy+Xc=Tc IF Iht=1 THEN Xt=Ya Ya=Ya/Xa Xa-Xt END IF X-(Xa-Xmin)-Sfx Y-(Ya-Ymin)-Sfy IF Y>100 OR Y<0 THEN 7220 PRINT "PA",X.Y."" HEXT I BEEP TNDIT """" 7210 7220 7230 7230 7240 7250 7250 7250 7250 7250 7270 7280 7290 7300 BEEF INPUT "WANT TO PLIT A LEAST-SOUARES LINE".IIs IF IIs-1 THEN PRINT "SM" Bb=(Npairs=Sxy=Sy)/(Npairs=Sx2=Sx 2) Aa=(Sy=Bb=Sx)/Npairs Aa=EXP(Aa) PRINTUSING "10X.""a = "".Z.0DE":Aa PRINTUSING "10X.""a = "".Z.4DE":Bb PRINTUSING "10X.""a = "".Z.4DE":Bb PRINTUSING "10X.""a = "".Z.4DE":Bb PRINTUSING "10X.""a = "".Z.4DE":Aa PRINTUSING "10X.""a = "".Z.4DE":A PRINTUSING "10X.""a = "".Z.4DE":A PRINTUSING "10X.""a = "".Z.4DE":A PRINTUSING "10X.""a = "".Z.4DE":A PRINTUSING "10X.""a = "A= PRINTUSING "10X.""a = "A PRINTUS Bb=(Npairs+Sxy-Sy+Sx)/(Npairs+Sx2-Sx 2) 7330 7340 PRINT THUR ALL FI NEXT Xa PRINT "PU" END IF ASSIGN #File TD + 7430 GDTO 6650 END IF PPINT "PU SM" BEEP 7480 DELF INPUT "WANT TO PLOT NUSSELT LINE (1=Y.0=N)?".Inp IF Inp=0 THEN 7620 PEEP 7510 7520 7530 INPUT "ENTER SLOPE FOR CONSTANT IN HNU VS 0", SI BEEP

Separate District Con Substant IN BHg as a Config FOR Xa Amin 10 Amax s125 Astes/50 fareary:\*Si\*Ka\*Xa, 5.03 X\*(Xa-Xmin)\*Sfx Y\*(Ya-Ymin)\*Sfy PRINT "PA",X,Y,"PD" HEXT Xa 758.: 2520 7580 7590 7600 7610 7620 7630 NEXT Xa PRINT "PU PA 0.0" DEEP INPUT "LIKE TO DRAW A STRAIGHT LINE (1=Y.0=N)?".Go\_on IF Go\_on=1 THEN EEEP 7640 7650 7660 7670 INPUT "ENTER THE SLOPE".S1 7680 7690 7700 7710 7720 BEEP INPUT "ENTER THE INTERCEPT".Ac FOR Xa=Xmin TU Xmax STEP (Xmax-Xmin) fa=Ac+S1+Xa Y=(Ya=Ymin)\*Sfy X=(Xa=Xmin)\*Sfx IF Y<0 THEN Xam=(Ymin=Ac)/S1 BEEP 7730 7740 7750 7760 7760 7770 7780 7790 X=(Xam-Xmin)=Sfx 1=0 END IF IF Y>100 THEN Xama(Ymax-Ac)/Sl 7800 7810 7820 7830 X=(Xam=(min)=Sfx Y=100 7840 7850 7360 INPUT "WANT TO PLOT A POLYNOMIAL (I=Y.0=N)?".Go\_on IF Go\_on=1 THEN BEEP 7970 1380 7890 7300 7910 INPUT "ENTER THE ORDER OF POLYNOMIAL".M THE SECTION AND A CONTRACT OF THE SECTION OF THE SE 7920 7930 2940 7950 7960 7970 NEXT NEXT J Ya=C(0) FOR Xa=Xmin TO Xmax STEP Xstep/10 FOR J=1 TO M '<u>9</u>80 1990 Ya=Ya+C(J)=Xa J NEXT J Y=(Ya-Ymin)=Sfy :000 3010 020 Y+(Ya-Ymin)+Sfx IF Y<0 THEN Y+0 IF Y>100 THEN GOTO 8070 FRINT "PA".X.Y."PD" 3030 2140 3050 :050 NEXT Xa END IF PRINT "PU PA 0.0 SPO" 3070 1080 1090 1100 SUBEND

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