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LONGITUDINAL VIBRATIONS OF RODS OF FINITE LENGTH WITH RADIAL DEFORMATION

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20. ABSTRACT (CONT'D)

the set of two partial differential equations is recapitulated together with appropriate boundary conditions. For vibration problems, two sets of eigenvalue problems are formulated to satisfy the simultaneous partial differential equations and the homogeneous boundary conditions. Suitable parameters are defined to describe the dispersion relations. These dual eigenvalue matrix equations are then solved numerically. For an infinite rod, a dispersion relation of frequency versus wave number which contains an imaginary branch has been obtained. The free vibration problem of a fixedfixed Mindlin-Herrmann rod has been solved. The numerical values of six (6) lowest frequencies, the associated wave numbers and mode shapes are tabulated for three different slenderness ratios.

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TABLE OF CONTENTS

F,

	Page
ACKNOWLEDGEMENTS	ii
LIST OF SYMBOLS	iii
INTRODUCTION	1
GOVERNING EQUATIONS	2
SOLUTION FORMULATIONS FOR RODS OF INFINITE LENGTH	4
NUMERICAL RESULTS FOR RODS OF INFINITE LENGTH	8
SOLUTION FORMULATIONS OF RODS OF FINITE LENGTH	9
NUMERICAL RESULTS ON VIBRATIONS OF FINITE RODS WITH AXIAL AND RADIAL MOTIONS	15
REFERENCES	16

TABLES

I.	VIBRATION SOLUTIONS TO A FINITE MINDLIN-HERRMANN ROD WITH	17
	FIXED-FIXED END CONDITIONS ($v = 0.29$, Slenderness Ratio =	
	0.05)	

- II. VIBRATION SOLUTIONS TO A FINITE MINDLIN-HERRMANN ROD WITH 18 FIXED-FIXED END CONDITIONS (v = 0.29, Slenderness Ratio = 0.10)
- III. VIBRATION SOLUTIONS TO A FINITE MINDLIN-HERRMANN ROD WITH 19 FIXED-FIXED END CONDITIONS (v = 0.29, Slenderness Ratio = 0.20)

LIST OF ILLUSTRATIONS

- Longitudinal wave velocities of first mode of motion according 20 to Mindlin-Herrmann Rod Theory (same as curves la and l'a in Figure 1, Reference 1).
- 2. Effect of coupling through Poisson's Ratio, on wave velocities 21 of axial and radial modes (same as Figure 2, Reference 1).
- 3. Frequency versus real and imaginary "wave number" in an 22 infinite Mindlin-Herrmann Rod.

LIST OF SYMBOLS

а	:	radius of the rod
w,u	:	displacements in z and r-directions, respectively
λ,μ	:	Lame´ elastic constants
ρ	:	density
k,kl	:	correction factors
Y	:	wave number
ω	:	frequency
L	:	wave length
c	:	phase velocity (= ω/γ)
cs	:	shear wave velocity (= $\sqrt{\mu/\rho}$)
٤	:	length of the rod
ω*	:	dimensionless frequency
γ *	:	dimensionless wave number
a *	:	slenderness ratio
β -	:	coefficients of the associated eigenfunctions for w

 $\hat{\beta}$: coefficients of the associated eigenfunctions for u

. . . .

This report investigates the free vibration problem of a rod of finite length which models the axial as well as radial modes of motion. The formulation is based on a rod model of infinite length developed by Mindlin and Herrmann in 1951 (ref 1). The differential equations, boundary conditions, and the associated energy principle have all been developed in the original paper as well as dispersion relations (relations between wave number and velocity) for a rod of infinite length. Subsequently, Herrmann presented a solution formulation for free and forced vibration of rods of finite length (ref 2). No numerical results were given there, however. Miklowitz has obtained formed solutions to forced vibration problems of rods of semiinfinite and finite length by the use of Laplace transforms (ref 3). He has also obtained numerical results for the earlier formulations (ref 4). Nevertheless, no specific numerical data were provided for free vibrations of a finite rod. Since the free vibration information is fundamental to all linear and nonlinear wave propagation phenomena, our first step is to study this aspect for a Mindlin-Herrmann rod of finite length.

¹R. D. Mindlin and G. Herrmann, "A One-Dimensional Theory of Compressional Waves in an Elastic Rod," Proceedings of the First U.S. National Congress of Applied Mechanics, 1950, pp. 187-191.

²G. Herrmann, "Forced Motions of Elastic Rods," <u>Journal of Applied Mechanics</u>, September 1954, pp. 221-224.

 ³J. Miklowitz, "Travelling Compressional Waves in an Elastic Rod According to the More Exact One-Dimensional Theory," Proceedings of the Second U.S. National Congress of Applied Mechanics, June 1954, ASME, 1955, pp. 176-186.
 ⁴J. Miklowitz, "The Propagation of Compressional Waves in a Dispersive Elastic Rod," June 1957, pp. 231-239.

First, the governing partial differential equations and pertinent variables of the approximate theory of the rod are recapitulated. With the original equations, an eigenvalue problem based on a wave motion is formulated. The characteristic equation of this eigenvalue problem is shown to be identical to the one in Reference 1 in terms of phase velocity and wave number. However, the same equation can be written in terms of dimensionless frequency and wave number which is also allowed to assume imaginary values. Then, the dispersion relation of frequency versus wave number reveals another imaginary branch similar to the one corresponding to Timoshenko beam theory (ref 5). This new dispersion relation is shown here together with other results identical to the original paper by Mindlin and Herrmann. Next, solutions for finite rods are treated. As a first example, we take a rod with fixed-fixed ends. In order to satisfy these end conditions, another eigenvalue problem is established. This produces a characteristic equation for those particular frequencies so that free vibrations under the given boundary conditions are possible. The numerical results are presented; and among infinite number of discrete frequencies, several of the first lowest are recorded. Corresponding to each, the various modes (consisting of sine, cosine, hyperbolic sine, and cosine functions) and relative amplitudes are also presented in the last section.

¹R. D. Mindlin and G. Herrmann, "A One-Dimensional Theory of Compressional Waves in an Elastic Rod," Proceedings of the First U.S. National Congress of Applied Mechanics, 1950, pp. 187-191.

⁵S. P. Timoshenko, "On the Correction for Shear of the Differential Equation for Transverse Vibrations of Prismatic Bars," <u>Philosophical Magazine</u>, Series 6, Vol. 41, 1951, pp. 744-746.

GOVERNING EQUATIONS

It will be convenient here to recapitulate the full set of equations of the Mindlin-Herrmann rod model (ref 1). The displacement equations of motions are:

$$a^{2}(\lambda+2\mu)w'' + 2a\lambda\mu' + 2aZ = pa^{2}w \qquad ...$$

$$a^{2}k^{2}u'' - 8k_{1}^{2}(\lambda+\mu)u - 4ak_{1}^{2}\lambda w' + 4aR = \rho a^{2}u \qquad (1)$$

where w = w(z,t) is the displacement component in z-direction; u = u(z,t) so that ru(z,t)/a is the displacement component in r-direction; z, r denote the axial and radial coordinates, respectively; t, the time. A prime (') denotes partial differential with respect to z and a dot (.), the same with respect to t. λ , μ are Lame⁻ elastic constants, ρ is the density of the rod material, and a is the radius. R and Z denote the radial and axial components, respectively, of the traction on the cylindrical surface of the rod. k and k₁ are correction factors. They are introduced so that the dispersion relations from this approximate theory will better match the exact solution of an infinite rod. Their usage is discussed fully in the original paper by Mindlin and Herrmann (ref 1).

The stress-displacement relations are

$$2P_{r} = 2P_{\theta} = k_{1}^{2} [2a(\lambda + \mu)u + a^{2}\lambda w']$$

$$2P_{z} = 2a\lambda u + a^{2}(\lambda + 2\mu)w'$$

$$40 = k^{2}a^{2}\mu u'$$

$$(2)$$

¹R. D. Mindlin and G. Herrmann, "A One-Dimensional Theory of Compressional Waves in an Elastic Rod," Proceedings of the First U.S. National Congress of Applied Mechanics, 1950, pp. 187-191. where P_r , P_θ , P_z , and Q are the averaged stresses which are related to the Cauchy stress components σ_{rr} , etc., by the following relations:

$$P_{r} = \int_{0}^{a} \sigma_{rr} r dr; \qquad P_{\theta} = \int_{0}^{a} \sigma_{\theta\theta} r dr$$

$$P_{z} = \int_{0}^{a} \sigma_{zz} r dr; \qquad Q = \int_{0}^{a} \frac{\sigma_{rz}}{a} r^{2} dr$$
(3)

Equation (2) will be used to obtain solutions of finite rods if stress boundary conditions at the ends are given.

In the following discussion, we shall assume that the cylindrical surface of the rod is free of stress. Hence

$$\mathbf{R} = \mathbf{Z} = \mathbf{0} \tag{4}$$

and Eq. (1) becomes:

$$a^{2}(\lambda+2\mu)w'' + 2a\lambda u' = \rho a^{2}w ...$$

$$a^{2}k^{2}u'' - 8k_{1}^{2}(\lambda+\mu)u - 4ak_{1}^{2}\lambda w' = \rho a^{2}u$$
(1')

SOLUTION FORMULATIONS FOR RODS OF INFINITE LENGTH

To solve Eq. (1'), we divide through by ρa^2 and obtain:

$$\left(\frac{\lambda+2\mu}{\rho}\right)\mathbf{w}'' + \frac{2\lambda}{\rho a}\mathbf{u}' = \mathbf{w}$$

$$\left(\frac{k^{2}\mu}{\rho}\right)\mathbf{u}'' - \frac{8k_{1}^{2}(\lambda+\mu)}{\rho a^{2}}\mathbf{u} - \frac{4k_{1}^{2}\lambda}{\rho a}\mathbf{w}' = \mathbf{u}$$

$$\left(\frac{k^{2}\mu}{\rho}\right)\mathbf{u}'' - \frac{8k_{1}^{2}(\lambda+\mu)}{\rho a^{2}}\mathbf{u} - \frac{4k_{1}^{2}\lambda}{\rho a}\mathbf{w}' = \mathbf{u}$$

$$\left(\frac{k^{2}\mu}{\rho a}\right)\mathbf{u}'' + \frac{k_{1}^{2}\lambda}{\rho a^{2}}\mathbf{u} + \frac{k_{1}^{2}\lambda}{\rho a}\mathbf{w}' = \mathbf{u}$$

$$\left(\frac{k^{2}\mu}{\rho a}\right)\mathbf{u}'' + \frac{k_{1}^{2}\lambda}{\rho a^{2}}\mathbf{u} + \frac{k_{1}^{2}\lambda}{\rho a}\mathbf{w}' = \mathbf{u}$$

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$$\left(\frac{k^{2}\mu}{\rho a}\right)\mathbf{u}'' + \frac{k_{1}^{2}\lambda}{\rho a^{2}}\mathbf{w}' = \mathbf{u}$$

$$\left(\frac{k^{2}\mu}{\rho a}\right)\mathbf{u} + \frac{k_{1}^{2}\lambda}{\rho a^{2}}\mathbf{w}' = \mathbf{u}$$

Let

$$\begin{cases} i\gamma z_{e}-i\omega t \\ w(z,t) = Ae \\ i\gamma z_{e}-i\omega t \\ u(z,t) = Be \end{cases}$$

$$(6)$$

and substitute Eq. (6) into Eq. (5). One has

$$\begin{bmatrix} -(\frac{\lambda+2\mu}{\rho})\gamma^{2} + \omega^{2}]A + i\gamma(\frac{2\lambda}{\rho a})B = 0 \\ -i\gamma(\frac{4k_{1}^{2}\lambda}{\rho a})A + [\frac{k^{2}\mu}{\rho}\gamma^{2} - \frac{8k_{1}^{2}(\lambda+\mu)}{\rho a^{2}} + \omega^{2}]B = 0 \end{bmatrix}$$
(7)

Divide Eq. (7) through by γ^2 :

$$\begin{bmatrix} -\frac{(\lambda+2\mu)}{\rho} + \frac{\omega^{2}}{\gamma^{2}} \end{bmatrix} A + i(\frac{2\lambda}{\rho\gamma_{a}})B = 0 \\ -i(\frac{4k_{1}^{2}\lambda}{\rho\gamma_{a}})A + \begin{bmatrix} -\frac{k^{2}\mu}{\rho} - \frac{8k_{1}^{2}(\lambda+\mu)}{\rho\gamma^{2}a^{2}} + \frac{\omega^{2}}{\gamma^{2}} \end{bmatrix} B = 0 \end{bmatrix}$$
(7')

From Eq. (7'), one obtains the following characteristic equation:

$$\left(\frac{\lambda+2\mu}{\rho}-\frac{\omega^{2}}{\gamma^{2}}\right)\left(\frac{k^{2}\mu}{\rho}+\frac{8k_{1}^{2}(\lambda+\mu)}{\rho\gamma^{2}a^{2}}-\frac{\omega^{2}}{\gamma^{2}}\right)+\frac{8k_{1}^{2}\lambda^{2}}{\rho^{2}\gamma^{2}a^{2}}=0$$
(8)

and the amplitude ratio

$$\alpha = \frac{B}{A} = -\left[-\frac{\lambda + 2\mu}{\rho} + \frac{\omega^2}{\gamma^2}\right] / \left[1\left(\frac{2\lambda}{\rho\gamma_a}\right)\right]$$
$$= + \left[1\left(\frac{-4k_1^2\lambda}{\rho\gamma_a}\right)\right] / \left[-\frac{k^2\mu}{\rho} - \frac{8k_1^2(\lambda + \mu)}{\rho\gamma^2a^2} + \frac{\omega^2}{\gamma^2}\right]$$
(9)

where Eq. (8) is exactly the same as Eq. (22) in Mindlin-Herrmann's paper of 1951 (ref 1).

It would be convenient for the present analysis to introduce some dimensionless parameters. But first, let

¹R. D. Mindlin and G. Herrmann, "A One-Dimensional Theory of Compressional Waves in an Elastic Rod," Proceedings of the First U.S. National Congress of Applied Mechanics, 1950, pp. 187-191.

l = length of the rod

L = wave length

and hence the wave number can be defined as

$$\gamma = 2\pi/L \tag{10}$$

Now, define

$$a^{\star} = a/\ell , \quad \gamma^{\star} = \gamma \ell = 2\pi \ell/L$$

$$(\omega^{\star})^{2} = \omega^{2} a^{2}/(\mu/\rho) , \quad \lambda^{\star} = \lambda/\mu$$
(11)

If, from Eq. (6), we define wave (phase) velocity as

$$c = \omega/\gamma \tag{12}$$

and

$$c_s = (\mu/\rho)^{1/2}$$

one has the relation

$$\frac{c^{2}}{c_{s}^{2}} = \frac{\omega^{2}}{----} = \frac{\omega^{2}a^{2}}{\gamma^{*}a_{a}^{2}} = \frac{\omega^{*}}{\gamma^{*}a_{a}^{2}}$$

$$\frac{\mu}{\rho} \gamma^{2} = \frac{\mu}{\rho} \gamma^{2}a^{2}$$
(13)

With Eqs. (11) and (13), Eq. (8) becomes the following when divided through by $\mu^2/\rho^2 = c_s^2$:

$$(\lambda^{*} + 2 - \frac{\omega^{*2}}{\gamma^{*2}a^{*2}})(k^{2} + \frac{8k_{1}^{2}(\lambda^{*}+1)}{\gamma^{*2}a^{*2}} - \frac{\omega^{*2}}{\gamma^{*2}a^{*2}}) - \frac{8k_{1}^{2}\lambda^{*2}}{\gamma^{*2}a^{*2}} = 0$$
(14)

Equation (9) can also be written as

$$\alpha = \frac{B}{A} = -i(\lambda^{*} + 2 - \frac{\omega^{*2}}{\gamma^{*2}a^{*2}})/(\frac{2\lambda^{*}}{\gamma^{*}a^{*}}) = \frac{4k_{1}^{2}\lambda^{*}}{\gamma^{*}a^{*}} = \frac{4k_{1}^{2}\lambda^{*}}{\gamma^{*}a^{*}} + \frac{8k_{1}^{2}(\lambda^{*}+1)}{\gamma^{*2}a^{*2}} - \frac{\omega^{*2}}{\gamma^{*2}a^{*2}})$$
(15)

Now, we shall solve γ^*a^* for any given ω^* in Eq. (14). Let

$$\begin{array}{c} x = \omega^{\ast 2} \\ 1 \\ y = \frac{1}{\gamma^{\ast 2} a^{\ast 2}} \end{array}$$
 (16)

In terms of x and y, Eq. (14) becomes

$$(\lambda + 2 - xy)[k^{2} + 8k_{1}^{2}(\lambda + 1)y - xy] - 8k_{1}^{2}\lambda + 2y = 0$$

Or, upon expanding,

$$[x^{2} - 8k_{1}^{2}(\lambda + 1)x]y^{2} - [(k^{2} + \lambda + 2)x - 8k_{1}^{2}(3\lambda + 2)]y + k^{2}(\lambda + 2) = 0$$
(17)

Or,

$$Ay^2 - By + C = 0$$
 (18)

with

$$A = x^{2} - 8k_{1}^{2}(\lambda * + 1)x$$

$$B = (k^{2} + \lambda * + 2)x - 8k_{1}^{2}(3\lambda * + 2)$$

$$C = k^{2}(\lambda * + 2)$$
(19)

Thus, one can write

$$y_{1,2} = \frac{B \pm \sqrt{B^2 - 4AC}}{2A}$$
(20)

Before we proceed to the actual computations, some observations on the sign of A, B, and B^2 - 4AC should be helpful.

1. A = x[x - $8k_1^2(\lambda + 1)$], and, since x = ω^{*2} is always positive or zero, for $\omega^* \neq 0$, A will be negative, zero, or positive depending on whether x is less, equal, or greater than $8k_1^2(\lambda^*+1)$:

$$A \stackrel{\langle}{=} 0 \text{ if } x \stackrel{\langle}{=} 3k_1^2(\lambda + 1) \sim 16$$

2.

$$B = (k^{2} + \lambda^{*} + 2)x - 8k_{1}^{2}(3\lambda^{*} + 2)$$
$$= (k^{2} + \lambda^{*} + 2)[x - \frac{8k_{1}^{2}(3\lambda^{*} + 2)}{k^{2} + \lambda^{*} + 2}]$$

Hence

$$B \stackrel{<}{=} 0 \text{ if } x \stackrel{<}{=} \frac{8k_1^2(3\lambda^{+2})}{k^2 + \lambda^{+2}} \sim 10$$

where the symbol ~ indicates "in the neighborhood when k, k₁ and λ^* are recognized to have an order of magnitude of unity."

3. Finally, let us consider $B^2 - 4AC$:

$$B^{2} = (k^{2}+\lambda *+2)^{2}x^{2} + 64k_{1}^{4}(3\lambda *+2)^{2} - 16k_{1}^{2}(3\lambda *+2)(k^{2}+\lambda *+2)x$$

$$4AC = 4[x^{2} - 8k_{1}^{2}(\lambda *+1)x]k^{2}(\lambda *+2) = 4k^{2}(\lambda *+2)x^{2} - 32k^{2}k_{1}^{2}(\lambda *+1)(\lambda *+2)x$$

$$= 4k^{2}(\lambda *+2)x^{2} - 32k_{1}^{2}(\lambda *^{2}+3\lambda *+2)k^{2}$$

$$B^{2} - 4AC = [k^{2} - (\lambda *+2)]^{2}x^{2} + 64k_{1}^{4}(3\lambda *+2)^{2}$$

$$- 16k_{1}^{2}x[3\lambda *+2)(\lambda *+2) - k^{2}(2\lambda *^{2}+3\lambda *+2)]$$

- a. For x being very large compared with unity, the first term dominates, hence, $B^2 4AC > 0$.
- b. For x being very small compared with unity, the second term dominates, and hence, $B^2 4AC > 0$ again.
- c. For x being in the neighborhood of unity, $B^2 4AC \sim 4 + 1600 128 =$ 1476 > 0.

Hence, in most likelihood, $B^2 - 4AC > 0$ for all values of x. But we shall examine this carefully in actual computations.

NUMERICAL RESULTS FOR RODS OF INFINITE LENGTH

The solution formulations of the previous section are now carried out for specific numerical parameters. The results are shown in Figures 1, 2, and 3. In Figure 1, two dispersion curves (i.e., c/c vs. a/L) reported in Reference 1 are reproduced here for the purpose of numerical verifications. As in Reference 1, we have set v = 0.29. The curve la corresponds to $k_1 = 1$ and

¹R. D. Mindlin and G. Herrmann, "A One-Dimensional Theory of Compressional Waves in an Elastic Rod," Proceedings of the First U.S. National Congress of Applied Mechanics, 1950, pp. 187-191.

k = 0.9258 which is obtained from Eq. (25) in Reference 1; the curve 1'a corresponds to the same k, but k_1 is obtained by Eq. (28) of the same reference. In Figure 2, the coupling effect of the axial and radial modes through Poisson's ratio given in Reference 1 is also reproduced here. As shown in Figure 2, there exist two velocities for each wave number a/L in its full range. However, it will be realized that the full range of frequency (ω *) is not covered in Figure 2. Since ω * is related to the velocity c/c_s by Eq. (13), it is necessary to solve ω^* for given $a/L = 2\pi\gamma^*a^*$ in the characteristic equation (14). Hence, another dispersion relation is obtained and is shown in Figure 3 for ω^* vs. $2\pi\gamma^*a^*$ curves. Evidently, in this figure, full range of frequency is covered. It is observed that for ω^* smaller than a value near 4.3, one root of γ *a* is imaginary, while the other is real. The eigenfunctions corresponding to the imaginary root are hyperbolic sine and cosine functions. They, together with the sine and cosine functions associated with the real root of γ *a*, will be needed for vibration solutions for a given set of end conditions of a finite rod.

SOLUTION FORMULATIONS OF RODS OF FINITE LENGTH

From solutions of rods of infinite length, one observes that there are four fundamental solutions in the form of sines and cosines or hyperbolic sine and cosine functions corresponding to any given frequency ω . Now, we wish to find a particular ω for a given specific length of the rod and for a given set of boundary conditions. It is also observed that the fundamental solutions of

¹R. D. Mindlin and G. Herrmann, "A One-Dimensional Theory of Compressional Waves in an Elastic Rod," Proceedings of the First U.S. National Congress of Applied Mechanics, 1950, pp. 187-191.

 $\boldsymbol{\omega}$ and \boldsymbol{u} are matched as follows:

$$w = \overset{=}{\omega_{1}} \cos \gamma * z * e^{-i\omega t} \quad \text{with} \quad u = \overset{=}{u_{1}} \sin \gamma * z * e^{-i\omega t}$$

$$w = \overset{=}{\omega_{2}} \sin \gamma * z * e^{-i\omega t} \quad \text{with} \quad u = \overset{=}{u_{2}} \cos \gamma * z * e^{-i\omega t}$$

$$w = \overset{=}{\omega_{3}} \cosh \gamma * z * e^{-i\omega t} \quad \text{with} \quad u = \overset{=}{u_{3}} \sinh \gamma * z * e^{-i\omega t}$$

$$w = \overset{=}{\omega_{4}} \sinh \gamma * z * e^{-i\omega t} \quad \text{with} \quad u = \overset{=}{u_{4}} \cosh \gamma * z * e^{-i\omega t}$$

$$(21)$$

where

and

$$\alpha_{1} = -\left[-\left(\lambda^{*}+2\right) + \frac{\omega^{*2}}{\gamma^{*2}a^{*2}}\right] / \left(\frac{2\lambda^{*}}{\gamma^{*}a^{*}}\right)$$

$$\alpha_{2} = -\alpha_{1}$$

$$\alpha_{3} = \left[\left(\lambda^{*}+2\right) + \frac{\omega^{*2}}{\gamma^{*2}a^{*2}}\right] / \left(\frac{2\lambda^{*}}{\gamma^{*}a^{*}}\right)$$

$$\alpha_{4} = -\alpha_{3}$$

$$(23)$$

From the dispersion relations one observes that $(\gamma^*a^*)^2$ always has two roots for any given ω^* : either (1) both of these roots are real, or, (2) one is real and the other, imaginary.

For case (1), the general solution has the form

 $w(z,t) = (w_1 \cos \gamma_1 * z^* + w_2 \sin \gamma_1 * z^* + w_3 \cosh \gamma * z^* + w_4 \sinh \gamma * z^*)e^{-i\omega t} (24)$ $u(z,t) = (u_1 \sin \gamma_1 * z^* + u_2 \cos \gamma_1 * z^* + u_3 \sinh \gamma * z^* + u_4 \cosh \gamma * z^*)e^{-i\omega t} (25)$

As our first example, the boundary conditions of a rod with both ends fixed will be considered:

$$\begin{array}{c} w(0,t) = w(l,t) = 0 \\ u(0,t) = u(l,t) = 0 \end{array}$$
 (26)

Using Eq. (26) for Eqs. (24) and (25), one has, respectively

$$\vec{w}_{1} + 0 + \vec{w}_{3} + 0 = 0$$

$$\vec{w}_{1} \cos \gamma_{1} + \vec{w}_{2} \sin \gamma_{1} + \vec{w}_{3} \cos \gamma_{3} + \vec{w}_{4} \sin \gamma_{3} = 0$$

$$0 + \alpha_{2} \vec{w}_{2} + 0 + \alpha_{4} \vec{w}_{4} = 0$$

$$\vec{u}_{1} = \vec{u}_{1} + \vec{u}_{2} \vec{w}_{2} + 0 + \alpha_{4} \vec{w}_{4} = 0$$

$$\vec{u}_{1} = \vec{u}_{1} + \vec{u}_{2} \vec{w}_{2} + 0 + \alpha_{4} \vec{w}_{4} = 0$$

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$$\vec{u}_{2} = \vec{u}_{1} + \vec{u}_{2} \vec{w}_{2} + 0 + \alpha_{4} \vec{w}_{4} = 0$$

$$\vec{u}_{1} = \vec{u}_{1} + \vec{u}_{2} \vec{w}_{2} + 0 + \alpha_{4} \vec{w}_{4} = 0$$

$$\alpha_1 w_1 \sin \gamma_1^* + \alpha_2 w_2 \cos \gamma_1^* + \alpha_3 w_3 \sin \gamma_3^* + \alpha_4 w_4 \cos \gamma_3^* = 0$$

with

$$\alpha_{1} = -\alpha_{2} = -[-(\lambda^{*}+2) + \frac{\omega^{*2}}{\gamma_{1}^{*2}a^{*2}}]/(\frac{2\lambda^{*}}{\gamma_{1}^{*}a^{*}})$$

$$\alpha_{3} = -\alpha_{4} = -[-(\lambda^{*}+2) + \frac{\omega^{*2}}{\gamma_{3}^{*2}a^{*2}}]/(\frac{2\lambda^{*}}{\gamma_{3}^{*}a^{*}})$$
(28)

and

with

$$a_{3} = -a_{4} = [(\lambda^{*}+2) + \frac{\omega^{*2}}{\gamma^{*2}a^{*2}}]/(\frac{2\lambda^{*}}{\gamma_{3}a^{*}})$$
(30)

while α_1 and α_2 are the same as above.

For a nontrivial solution of Eq. (27), one has

$$\Delta_{1} = \begin{bmatrix} 1 & 0 & 1 & 0 \\ \cos \gamma_{1}^{*} & \sin \gamma_{1}^{*} & \cos \gamma_{3}^{*} & \sin \gamma_{3}^{*} \\ 0 & \alpha_{2} & 0 & \alpha_{4} \\ \alpha_{1} \sin \gamma_{1}^{*} & \alpha_{2} \cos \gamma_{1}^{*} & \alpha_{3} \sin \gamma_{3}^{*} & \alpha_{4} \cos \gamma_{3}^{*} \end{bmatrix} = 0$$

$$\Delta_{1} = \begin{bmatrix} \sin \gamma_{1}^{*} & \cos \gamma_{3}^{*} - \cos \gamma_{1}^{*} & \sin \gamma_{3}^{*} \\ \alpha_{2} & 0 & \alpha_{4} \\ \alpha_{2} \cos \gamma_{1}^{*} & \alpha_{3} \sin \gamma_{3}^{*} - \alpha_{1} \sin \gamma_{1}^{*} & \alpha_{4} \cos \gamma_{3}^{*} \end{bmatrix}$$

$$\Delta_{1} = \alpha_{2} \sin \gamma_{3}^{*} (\alpha_{3} \sin \gamma_{3}^{*} - \alpha_{1} \sin \gamma_{1}^{*})$$

$$+ \alpha_{2}\alpha_{4} \sin \gamma_{1}^{*} (\alpha_{3} \sin \gamma_{3}^{*} - \alpha_{1} \sin \gamma_{1}^{*})$$

$$- \alpha_{4} \sin \gamma_{1}^{*} (\alpha_{3} \sin \gamma_{3}^{*} - \alpha_{1} \sin \gamma_{1}^{*})$$

$$- \alpha_{2}\alpha_{4} \cos \gamma_{3}^{*} (\cos \gamma_{3}^{*} - \cos \gamma_{1}^{*})$$

$$(31)$$

Or,

$$\Delta_1 = (\alpha_3 \sin \gamma_3^* - \alpha_1 \sin \gamma_1^*)(\alpha_2 \sin \gamma_3^* - \alpha_4 \sin \gamma_1^*) - \alpha_2 \alpha_4 (\cos \gamma_3^* - \cos \gamma_1^*)^2 = 0$$
(32)

For a nontrivial solution of Eq. (29):

$$\Delta_{2} = \begin{bmatrix} 1 & 0 & 1 & 0 \\ \cos \gamma_{1}^{*} & \sin \gamma_{1}^{*} & \cosh \gamma_{3}^{*} & \sinh \gamma_{3}^{*} \\ 0 & \alpha_{2} & 0 & \alpha_{4} \\ \alpha_{1} \sin \gamma_{1}^{*} & \alpha_{2} \cos \gamma_{1}^{*} & \alpha_{3} \sinh \gamma_{2}^{*} & \alpha_{4} \cosh \gamma_{3}^{*} \end{bmatrix} = 0 \quad (33)$$

0r,

$$\Delta_2 = (\overline{\alpha_3} \sinh \gamma_3^* - \alpha_1 \sin \gamma_1^*)(\alpha_2 \sinh \gamma_3^* - \overline{\alpha_4} \sin \gamma_1^*) - \overline{\alpha_2}\alpha_4(\cosh \gamma_3^* - \cos \gamma_1^*)^2 = 0$$
(34)

With Eqs. (33) and (34), one can write

$$\Delta = \Delta(\omega^*) = \frac{\Delta_1(\omega^*) \text{ if } \gamma^{*2} a^{*2} \text{ has two real roots}}{\Delta_2(\omega^*) \text{ if } \gamma^{*2} a^{*2} \text{ has one real and one imaginary root}}$$
(35)

Hence, for a given ω^* , one must first find two values for $\gamma^{*2}a^{*2}$ and α_1 , α_2 , α_3 , and α_4 (or α_3 and α_4). Then with given $a^* = a/\ell$, one obtains γ_1^* and γ_3^* . Thus one can plot (ω^*) with respect to ω^* and find $\overline{\omega^*}$'s at which $\Delta(\overline{\omega^*}) = 0$.

Finally in this section, we will write down the formulas for computing the relative amplitudes of various mode shapes. Equation (27) for case (1) can be written as

$$\begin{bmatrix} 0 & 1 & 0 \\ \sin \gamma_{1}^{*} & \cos \gamma_{3}^{*} & \sin \gamma_{3}^{*} \\ \alpha_{2} & 0 & \alpha_{4} \end{bmatrix} \begin{bmatrix} z \\ w_{2} \\ w_{3} \\ w_{4} \\ w_{4} \end{bmatrix} = -\begin{bmatrix} z \\ \cos \gamma_{1}^{*} \\ 0 \\ 0 \end{bmatrix} \begin{bmatrix} z \\ w \\ w \\ 0 \end{bmatrix} (36)$$

From Eq. (36), one has

$$= \frac{D_2}{D} = \frac{1}{D} = \frac{1}{w_1} = \frac{1}{\beta_2 w_1}$$

$$= \frac{D_3}{D} = \frac{1}{w_1} = \frac{1}{\beta_3 w_1}$$

$$= \frac{D_4}{D} = \frac{1}{w_1} = \frac{1}{\beta_4 w_1}$$

$$(37)$$

where

$$D = \begin{bmatrix} 0 & 1 & 0 \\ \sin \gamma_{1}^{*} & \cos \gamma_{3}^{*} & \sin \gamma_{3}^{*} \\ \alpha_{2} & 0 & \alpha_{4} \end{bmatrix} = \alpha_{2} \sin \gamma_{3}^{*} - \alpha_{4} \sin \gamma_{1}^{*}$$

$$D_{2} = \begin{bmatrix} -1 & 1 & 0 \\ -\cos \gamma_{1}^{*} & \cos \gamma_{3}^{*} & \sin \gamma_{3}^{*} \\ 0 & -1 & \alpha_{4} \end{bmatrix} = \alpha_{4}(\cos \gamma_{1}^{*} - \cos \gamma_{3}^{*})$$

$$D_{3} = \begin{bmatrix} 0 & -1 & 0 \\ \sin \gamma_{1}^{*} & -\cos \gamma_{1}^{*} & \sin \gamma_{3}^{*} \\ \alpha_{2} & 0 & 0 \end{bmatrix} = \alpha_{4} \sin \gamma_{1}^{*} - \alpha_{2} \sin \gamma_{3}^{*}$$

$$D_{4} = \begin{bmatrix} 0 & 1 & -1 \\ \sin \gamma_{1}^{*} & \cos \gamma_{3}^{*} & -\cos \gamma_{1}^{*} \\ \alpha_{2} & 0 & 0 \end{bmatrix} = \alpha_{2}(\cos \gamma_{3}^{*} - \cos \gamma_{1}^{*})$$
(38)

For case (2), the above equation can still be used with cos γ_3^* and sin γ_3^* replaced by cosh γ_3^* and sinh γ_3^* . Now, for u_1 , i = 1,2,3,4, we observe Eqs. (22), (23), (28), and (30), and write

$$u_{i} = a_{i}w_{i} = \beta_{i}w_{i}$$
, $i = 1, 2, 3, 4$ (39)

with

$$\vec{\beta}_{i} = \alpha_{i}\beta_{i}$$
, $i = 1, 2, 3, 4$ (40)

where $\beta_1 = 1$ when w_1 is used as the basis of normalization,

$$\beta_{1} = \frac{D_{1}}{D_{1}}$$
, $i = 2, 3, 4$
 $\beta_{1} = 1$ (41)

In Eq. (40), α_3 and α_4 will be used in place of α_3 and α_4 if the mode shape involves hyperbolic cosine and sine as stated in the earlier discussion.

NUMERICAL RESULTS ON VIBRATIONS OF FINITE RODS WITH AXIAL AND RADIAL MOTIONS

The formulations obtained in the previous section are now used for specific numerical examples. In conjunction with a fixed-fixed rod (i.e., w and u are zero at both ends), several lowest vibration frequencies and the associated eigenfunctions have been obtained. The results are tabulated in Tables I through III for slenderness ratio $a^* = a/\ell = 0.05$, 0.10, and 0.20, respectively. In these tables, β_1 , i = 1,2,3,4 are the coefficients for w(x,t) and β_1 for u(x,t). They are normalized with respect to the one with largest absolute value. For any given ω^* , a pair of γ^*a^* is obtained. These are indicated by γ_1^*a and γ_3^*a , respectively, as described in the previous section. If one of them (γ_1^*a) is imaginary, i.e., $\overline{\gamma_1}^*a = i\overline{\gamma_1}^*a$, then γ_1^*a is given and it is indicated by parentheses. The coefficients of the associated eigenfunctions (hyperbolic sine and cosine functions) are also indicated by parentheses. For example, in Table I for $a^* = a/\ell = 0.5$, both the lowest frequencies have one imaginary root each. For all higher ω^* 's, both roots of γ^*a^* are real.

REFERENCES

- R. D. Mindlin and G. Herrmann, "A One-Dimensional Theory of Compressional Waves in an Elastic Rod," Proceedings of the First U.S. National Congress of Applied Mechanics, 1950, pp. 187-191.
- G. Herrmann, "Forced Motions of Elastic Rods," <u>Journal of Applied</u> Mechanics, September 1954, pp. 221-224.
- 3. J. Miklowitz, "Travelling Compressional Waves in an Elastic Rod According to the More Exact One-Dimensional Theory," Proceedings of the Second U.S. National Congress of Applied Mechanics, June 1954, ASME, 1955, pp. 176-186.
- 4. J. Miklowitz, "The Propagation of Compressional Waves in a Dispersive Elastic Rod," June 1957, pp. 231-239.
- S. P. Timoshenko, "On the Correction for Shear of the Differential Equation for Transverse Vibrations of Prismatic Bar," <u>Philosophical</u> Magazine, Series 6, Vol. 41, 1951, pp. 744-746.

VIBRATION SOLUTIONS TO A FINITE MINDLIN-HERRMANN ROD WITH FIXED-FIXED END CONDITIONS (v = 0.29, slenderness Ratio = 0.05)TABLE I.

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Freq. Relative Magnitude of Modes	ω1* = 0.1575	w2* = 3.6600	w3* = 4.4813	ω4* = 4.6810	w5* = 4.8480	ω6 [*] = 5.1647
Y1*a*	(82.29)	(39.81)	15.58	25.31	31.03	39.29
β1	(-0.3380)	(0000.1)	-0.0018	0.3102	-0-0406	-0.1538
β2	(0.2707)	(0.5580)	0.0128	0.0588	0.1139	0.2496
¥3*a	1.962	51.36	68.71	73.41	77.47	85.31
β3	0.3380	-1.0000	-0-018	-0.0310	0.0406	0.1538
β4	0.2720	0.0000	1.0000	1.0000	1.0000	1.0000
Y1*a*	(82.29)	(18.96)	15.58	25.31	31.03	39.29
β ₁	(-0.1005)	(0000.0)	0.0248	-0-0913	0.0604	0.1001
β2	(0.0805)	(0000.0)	0.1770	0.1731	0.1696	0.1625
Y3*a*	1.962	51.36	68.71	73.41	77.47	85.31
- β3	1.0000	-0.1903	0.0003	-0.0054	0*0069	0.0250
64	-0.0805	0.000	-0.1770	-0.1731	-0.1696	-0.1625

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The numerical values in parentheses are associated with imaginary wave numbers and their hyperbolic sine and hyperbolic cosine (mode shape) functions. NOTE:

 TABLE II.
 VIBRATION SOLUTIONS TO A FINITE MINDLIN-HERRMANN ROD WITH FIXED-FIXED END CONDITIONS

 (v = 0.29, Slenderness Ratio = 0.10)

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Freq. Relative Magnitude of Modes	ω <u>1</u> * = 3.6600	w2* = 4.4391	wg* = 4.6787	ω4* = 4.7515	ω5* = 4.8801	w6 [*] = 4.°9863
Y1*a*	(16.91)	6.247	12.63	13.96	16.00	17.47
β	(1.0000)	-0.0022	0.0980	0.0842	0.0809	0.0897
B2	(0.2601)	0.0065	0.0584	0.0865	0.1259	0.1686
۲3*a	25.68	33.87	36.69	37.56	39.13	40.44
β3	-1.0000	0.0022	-0*0980	-0.0842	-0.0809	-0.0897
β₄	00000	, 1.0000	1.0000	1.0000	1.0000	1.0000
Y1*a*	(16.61)	6.247	12.63	13.96	16.00	17.47
β ₁	(00000.0)	0.0602	-0.2903	-0.1793	-0.1086	-0.0886
Β2	(00000.0)	0.1778	0.1732	0.1717	0.1689	0.1666
Y3*a*	25.68	33.87	36.69	37.56	39.13	40.44
- β3	-0.1903	0.004	-0.0170	-0.0145	-0.0137	-0.0149
В4	-0.0000	-0.1778	-0.1732	-0.1717	-0.1689	-0.1666

The numerical values in parentheses are associated with imaginary wave numbers and their hyperbolic sine and hyperbolic cosine (mode shape) functions. NOTE:

TABLE III. VIBRATION SOLUTIONS TO A FINITE MINDLIN-HERRMANN ROD WITH FIXED-FIXED END CONDITIONS (v = 0.29, Slenderness Ratio = 0.20)

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Freq. Relative						
Magnitude of Modes	ω ₁ * = 0.4156	w2* = 3.6600	w3* = 4.4396	ω ₄ * = 4.6785	ω5 * = 4.9000	w6* = 5.1331
Y1*a*	(20.48)	(9.952)	3.132	6.313	8.142	079-6
β1	(-0.8876)	(-0.1279)	-0.0092	0.2121	0.2383	0.3925
β2	(-0.1488)	(1.0000)	0.0066	0.0584	0.1336	0.2345
Y3*a	1.295	12.84	16.94	18.34	19.69	21.13
β3	0.8876	0.1279	0.0092	-0.2121	-0.2383	-0.3925
β₄	-0.0393	-0.0000	1.0000	1.0000	1.0000	1.0000
¥1*a*	(20.48)	(9.952)	3.132	6.313	8.142	9.640
β ₁	(-0.2645)	(0000.0)	0.2496	-0.6294	-0.3006	-0.2733
- β2	(-0.4433)	(0000.0)	0.1778	0.1732	0.1685	0.1633
Y3*a*	1.295	12.84	16.94	18.34	19.69	21.13
ء β3	1.0000	0.0243	0.0016	-0.0367	-0.0402	-0.0641
- 84	0.4433	0.000	-0.1778	-0.1732	-0.1685	-0.1633
					NA A A A A A A A A A A A A A A A A A A	

The numerical values in parentheses are associated with imaginary wave numbers and their hyperbolic sine and hyperbolic cosine (mode shape) functions. NOTE:







Figure 2. Effect of coupling through Poisson's Ratio, on wave velocities of axial and radial modes (same as Figure 2, Reference 1).



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Figure 3. Frequency versus real and imaginary "wave number" in an infinite Mindlin-Herrmann Rod.

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