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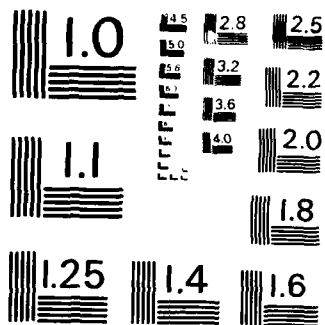
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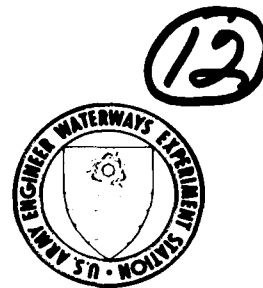
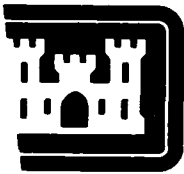
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TECHNICAL MEMORANDUM NO. 6-412

TENSILE CRACK EXPOSURE TESTS

Report 4

STATISTICAL ANALYSIS OF THE LONG-TERM DURABILITY OF SERIES "B" BEAMS

by

Henry T. Thornton, Jr.

Structures Laboratory

U. S. Army Engineer Waterways Experiment Station
P. O. Box 631, Vicksburg, Miss. 39180

March 1984

Report 4 of a Series

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Prepared for Office, Chief of Engineers, U. S. Army
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Under Civil Works Research Work Units 31132 and 31788

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| 20. ABSTRACT (Continue on reverse side if necessary and identify by block number) In November 1954, a long-term durability program was begun to determine the effects of severe natural weathering on reinforced concrete beams loaded to different stress levels and containing reinforcing steel with different types of bar deformations in either top-as-cast or bottom-as-cast positions. The beams were fabricated, cured, and loaded at the U. S. Army Engineer Waterways Experiment Station (WES) in 1954, then shipped to Eastport, Maine, and (Continued) | | |

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20. ABSTRACT (Continued). *DEG*

placed on the beach at the natural weathering exposure station on the south side of Treat Island in Cobscook Bay. The beams were subjected to twice daily tidal cycles exposing them to wetting under considerable head and drying to surface dry conditions. In addition, during the winter months, the beams were subjected to cycles of freezing and thawing with each tide when the air temperature was at or below 28° F (-2.2° C). The beams were inspected annually during the exposure period and evaluated by a team of inspectors rating the degree of deterioration. Nondestructive tests were also performed. Each year data on condition, percent velocity squared (V_L^2), and maximum crack width were collected.

The data which were generated from this study were coded and entered onto the WES IBM 4331 computer for subsequent analyses using the Statistical Analysis System (SAS). An evaluation of the results of these analyses indicates that:

a. Beams with steel in the bottom-as-cast position deteriorate at a slower rate than do beams with steel in the top-as-cast position for both A 305-50T and old-style deformation type, and beams with steel in the bottom-as-cast position exhibited smaller average maximum crack widths (significant at the 50,000-psi stress level).

b. A 305-50T type reinforcement bar deformation exhibited less severe degradation trends than old-style, and A 305-50T deformation type exhibited a significantly larger percent V^2 than did old-style deformation at the 50,000-psi stress level.

c. As stress levels increased, the conditions of the beams generally decreased and the degradation of percent V^2 increased. There were marked increases in maximum crack widths from the 40,000- to 50,000-psi stress levels for all positions and bar deformation types.

d. The more severe exposure conditions of the zero stress (control) beams, i.e., partially covered with sand where a state of higher saturation was maintained, probably affected some anomalous results. Also, the early failure of some 50,000-psi stress level beams containing reinforcement bars with old-style deformations and the subsequent loss of incriminating performance data affected some anomalous results.

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PREFACE

The statistical analysis reported herein was performed on data collected over the years from a test program planned by the Office, Chief of Engineers, in cooperation with the Reinforced Concrete Research Council of the American Society of Civil Engineers. The test program forms a part of Civil Works Research Work Unit 010401/31276 and was approved by the Office, Chief of Engineers, in 2nd indorsement, dated 17 Jan 1951, to basic letter, dated 7 Dec 1950, subject: "Reinforced Concrete Beams for Tensile Crack Exposure Tests," and has been conducted by the Concrete Technology Division (CTD), Structures Laboratory (SL), of the U. S. Army Engineer Waterways Experiment Station (WES).

The statistical analysis was performed as a part of Civil Works Research Work Unit 31132, "Field Exposure Durability Studies." Funds for the publication of this report were provided from Civil Works Research Work Unit 31788, "Special Studies for Civil Works Structural Engineering Problems," and from those made available for operation of the Concrete Technology Information Analysis Center (CTIAC). This is CTIAC Report No. 59. The report was prepared by Mr. Henry T. Thornton, Jr., under the general supervision of Messrs. Bryant Mather, Chief, SL, and John M. Scanlon, Chief, CTD.

Commander and Director of WES during publication of this report was COL Tilford C. Creel, CE. Technical Director was Mr. F. R. Brown.

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CONTENTS

| | <u>Page</u> |
|--|-------------|
| PREFACE | 1 |
| CONVERSION FACTORS, NON-SI TO SI (METRIC) UNITS OF MEASUREMENT . | 3 |
| PART I: PHYSICAL FACILITIES AND EXPOSURE CONDITIONS | 4 |
| PART II: SPECIMENS AND TEST PARAMETERS | 6 |
| Tensile Crack Specimens, Series B | 6 |
| Inspection and Testing | 6 |
| PART III: ANALYSIS SYSTEM AND PROCEDURE | 11 |
| Statistical Analysis System | 11 |
| Statistical Analysis of Variables Condition Rating, $\%v^2$, and Maximum Crack Width | 11 |
| Linear Models | 49 |
| PART IV: CONCLUSIONS | 50 |
| APPENDIX A: EXPOSURE RECORDS OF SPECIMENS | A1 |
| APPENDIX B: ANALYSIS OF VARIANCE | B1 |
| APPENDIX C: LINEAR REGRESSION ANALYSES | C1 |

CONVERSION FACTORS, NON-SI TO SI (METRIC)
UNITS OF MEASUREMENT

Non-SI units of measurement used in this report can be converted to SI (metric) units as follows:

| <u>Multiply</u> | <u>By</u> | <u>To Obtain</u> |
|--------------------------------|-----------|-----------------------------|
| Fahrenheit degrees | 5/9 | Celsius degrees or kelvins* |
| feet | 0.3048 | metres |
| inches | 25.4 | millimetres |
| pounds (force) per square inch | 6894.757 | pascals |
| feet per second | 0.3048 | metres per second |

* To obtain Celsius (C) temperature readings from Fahrenheit (F) readings, use the following formula: $C = (5/9)(F - 32)$. To obtain kelvin (K) readings, use: $K = (5/9)(F - 32) + 273.15$.

TENSILE CRACK EXPOSURE TESTS
STATISTICAL ANALYSIS OF THE LONG-TERM DURABILITY OF
SERIES "B" BEAMS

PART I: PHYSICAL FACILITIES AND EXPOSURE CONDITIONS

1. The ultimate test of the durability of concrete is its performance under the exposure conditions in which it is to serve. Although laboratory tests yield valuable indications of probable durability, the potential disrupting influences in nature are so numerous and variable that actual field exposures are highly desirable to assess the durability of concrete when exposed to natural weathering. An exposure station (Figure 1) located at Treat Island in Cobscook Bay near Eastport, Maine,

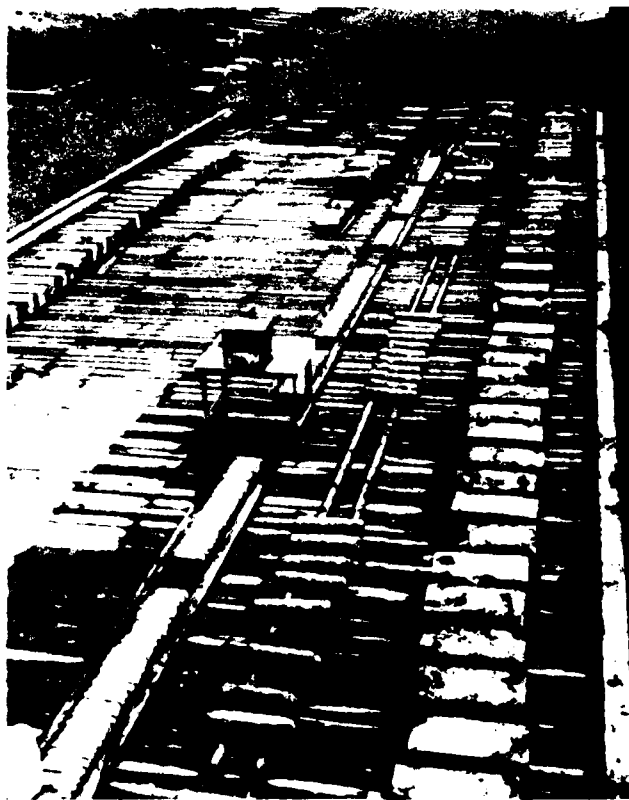


Figure 1. Exposure rack and beach area
where specimens are exposed

has been in use by the U. S. Army Corps of Engineers since 1936. Its location makes it ideal for exposing concrete and concreting materials to severe natural weathering. Its effect is to provide a natural field laboratory where no size limitation is placed on the exposed specimens. The specimens are installed at mean-tide elevation, and the alternating conditions of immersion of the specimens in seawater, then exposure to cold air, provide numerous cycles of freezing and thawing of the concrete during the winter. The effect of the relatively cool summers is to lessen, in general, autogenous healing and chemical reactions in the concrete.

2. In winter, the combination of air and water temperatures creates a condition in which specimens at the mean-tide elevation are thawed to a temperature of about 37° F* when covered with water and are frozen to temperatures as low as -10° F when exposed to air. A recording thermometer, the bulb of which is embedded in the center of a concrete specimen, records these temperatures. A cycle of freezing and thawing consists of the reduction of the temperature at the center of a concrete specimen to below 28° F and the subsequent rise to above 28° F. During an average winter, the specimens are subjected to over 100 cycles of freezing and thawing. In 26 winters, from 1953 to 1979, the number of annual cycles ranged from 71 to 185, with the average being 133.

3. There are currently 36 active research programs in progress at Treat Island involving the exposure of some 1700 concrete specimens. The annual testing and continuous monitoring of these programs yield valuable data on the durability and performance of concrete and concreting materials.

* A table of factors for converting non-SI units of measurement to SI (metric) units is presented on page 3.

PART II: SPECIMENS AND TEST PARAMETERS

Tensile Crack Specimens, Series B

4. In November 1954, 76 reinforced concrete beams were installed at half-tide elevation on the beach at Treat Island to compare the relative resistance to weathering of highly stressed, reinforced-concrete beams containing (a) reinforced bars deformed to conform to ASTM A 305-50T* and (b) bars with old-style deformations.

5. The beams were 7 ft 9 in. long and were made of air-entrained concrete with a nominal compressive strength of 2500 psi at 28 days age. All of the beams were reinforced with rail-steel bars; 38 beams contained reinforcement bars which conformed to ASTM A 305-50T and the remaining 38 beams contained reinforcement bars which conformed to the old-style deformations. Of these 76 reinforced beams, 64 of the beams were yoked and stressed by third-point loadings. The loadings ranged from 20,000 to 50,000 psi. The remaining 12 beams were designated as controls and were not loaded. Appendix A lists these specimens and gives their exposure records along with other pertinent information.

Inspection and Testing

6. From 1957 until 1979, the period over which the data for this analysis were collected, the relative resistance to weathering for each of these 76 beams was evaluated annually. Qualitative measurements pertaining to condition were recorded along with the quantitative measurements of pulse velocity and maximum crack width. Due to some midcourse corrections and the lack of concomitant data, the years 1966, 1967, 1973, and 1974 were excluded from this analysis.

Visual inspection and condition rating

7. All exposed specimens are inspected visually by the resident

* American Society for Testing and Materials, Book of ASTM Standards (issued in parts), revisions issued annually, Philadelphia, Pa.

contractor each week during the period that freezing-and-thawing cycles occur, usually October through March. The condition of each specimen is recorded on an inspection form which is forwarded to the laboratory along with the time-temperature history for that week. The inspection form is checked for noteworthy changes that may have occurred, and the number of freezing-and-thawing cycles that occurred during the week are taken from the time-temperature history.

8. During the summer of each year an inspection and testing team from the Structures Laboratory (SL), Waterways Experiment Station (WES), visits the exposure station for the purpose of performing the annual inspection and testing of all specimens by visual and other nondestructive methods. During this annual visit photographs are taken of all programs in progress with special emphasis on programs of particular interest at the time, and of any specimens exhibiting significant or ordinate deterioration.

9. At the same time during the data collection period (1957-1979), a four-man rating team consisting of representatives from WES and the Office, Chief of Engineers (OCE), and one or more from outside government completed condition rating forms on the Tensile Crack Concrete Beam program. Each beam received a score each year resulting from the combined rating forms (see example of form below). The opinions of the observers were remarkably concordant, with very few discrepancies noted over the years.

Inspection Sheets
Formal Inspection, Treat Island, Maine

Tensile Crack Exposure Tests Date _____

Instructions

1. Insert in column headed "No. of transverse cracks with spalling" the number of load cracks that have apparently chipped or spalled subsequent to formation when beams were loaded, that now have places in which a pencil can be inserted (about 1/4 in. wide).

2. Measure (Note) the total length of cracking, in inches, appearing over the reinforcing steel.

3. Measure the total length of reinforcement that can be seen through cracks, or that is exposed because concrete has spalled away from it.

4. Measure the total length of cracking bordered by iron stain from the crack.

5. Estimate the total area of visible horizontal and vertical surfaces of concrete that have scaled and make a check under the most appropriate heading on the rating sheet.

Note: Measure to $\pm 1/4$ in.

Scoring:

- a. Scoring will be done using a numerical system by others after the inspection.
- b. Score of zero indicates perfect condition.
- c. Light scaling scores 2, medium scaling 4, heavy scaling 8.
- d. Numerical score = sum of $4 \times$ number of spalled cracks + length of cracking over steel + $3 \times$ length of visible steel + length of cracking over steel bordering iron-stained areas + appropriate score for scaled area.

10. This score was then converted into a numerical condition rating. The general conversion scheme is shown below:

| <u>Condition</u> | <u>Score</u> | <u>Numerical Rating</u> |
|---|--------------|-------------------------|
| Negligible deterioration | 0 | 100 |
| Slight deterioration | 4 | 75 |
| More advanced deterioration | 104 | 50 |
| Advanced deterioration, usually with considerable exposure of reinforcing steel | 129 | 25 |
| Disintegrated, incapable of carrying load | 629 | 0 |

Pulse velocity tests

11. The concrete specimens are subjected also to ultrasonic pulse velocity tests in accordance with CRD-C 51* (ASTM C 597** each year during

* WES. 1949. Handbook for Concrete and Cement, with quarterly supplements, Vicksburg, Miss.

** Op cit.

exposure, unless their size, shape, or exposure condition prevents. The test instrument measures the time of travel of an ultrasonic pulse through a concrete specimen. From the travel time and the path length, values for pulse velocity (V) in the concrete are calculated. The square of the velocity thus determined is expressed as a percentage of the square of initial velocity obtained at installation (%V²). Example:

V_o = pulse velocity in a certain specimen at installation

V_t = pulse velocity in this same specimen at a later date

Therefore

$$\%V^2 \text{ (at time } t) = \frac{V_t^2}{V_o^2}$$

Since the square of the pulse velocity is related to the dynamic Young's modulus of elasticity, the %V² provides an alternate or supplementary parameter by which the progress of deterioration caused by natural weathering can be monitored. The initial velocity (V_o) of each beam was measured in 1954 so that the %V² comparison could be made in subsequent years. However, in 1955 and 1956 the velocities were not obtained. For this reason, and because the maximum crack width measurements were not initiated until 1957, the 1957 velocities were used as initial velocities and the statistical analysis was performed over the years 1957 to 1979.

Crack width measurements

12. Before shipment to the exposure station, beams of similar size, with similar stress in steel, and of similar concrete insofar as possible were paired and loaded with third-point flexural loading using spring and yoke devices. Nominal loads (stress in reinforcing steel) were 20,000, 30,000, 40,000, and 50,000 psi. Cracks developed in all of the loaded beams during loading. Beginning in 1957 the maximum width of cracks in the beams was measured annually using a measuring magnifier (least reading of 0.005 in.).

13. In 1963 after nine winters of exposure, comparisons were made of the effects of the variables of steel stress, position of steel at time of casting, and type of steel deformation, using condition rating,

$\%V^2$, and maximum crack width as quantitative measures. The results of these comparisons, as reported by Roshore* were as follows:

Based on condition rating--

The order of durability from most durable to least durable was zero stress, 20,000-, 30,000-, 40,000-, and 50,000-psi stress.

In 24 of the 45 comparable cases, the beams containing top-positioned steel exhibited greater durability than those containing bottom-positioned steel.

In 29 of the 50 comparable cases, beams containing steel meeting A 305-50T specifications exhibited better durability than those containing steel with old-style deformations.

Increase in crack width over time seemed to correlate with stress level, i.e., crack width increased with increasing stress in steel.

The changes in $\%V^2$ were highly variable from year to year and did not correlate well with results of visual inspections.

14. The objectives of this long-term study were multifaceted. Originally, the study was designed to evaluate the two types of reinforcement bars (A 305-50T and old-style deformations), the five levels of stress (0-, 20,000-, 30,000-, 40,000-, and 50,000-psi stress levels), and the position, as cast, of the steel within each concrete beam (top and bottom); however, subsequent to the initiation of this project and with respect to the constraints mandated by the experimental design, the interactions among these factors, i.e., the independence of factor combinations and the prediction of the measurable response, also became paramount to the successful interpretation of the relative resistance to weathering of these concrete beams.

* E. C. Roshore. 1964. "Tensile Crack Exposure Tests; Results of Tests of Reinforced Concrete Beams, 1955-1963," Technical Memorandum No. 6-412, Report 2, U. S. Army Engineer Waterways Experiment Station, CE, Vicksburg, Miss.

PART III: ANALYSIS SYSTEM AND PROCEDURE

Statistical Analysis System

15. The Statistical Analysis System (SAS) is a commercially available software package which operates on an IBM or IBM-compatible computer. SAS is one of the most reliable and up-to-date statistical packages available. Since WES has an IBM 4331 which is dedicated to SAS usage, the data generated from this study were keypunched and loaded onto a disk file associated with this minicomputer.

16. The analyses provided in this data report were generated by the MEANS procedure, the CORR procedure, and the ANOVA procedure. The MEANS procedure averages the replicates in each treatment combination. The CORR procedure generates the correlations between the quantitative variables, and the ANOVA procedure generates the analysis of variance tables and subsequent statistics.

Statistical Analysis of the Variables Condition Rating, %V², and Maximum Crack Width

17. The data which were generated from the long-term durability study consist of four descriptive factors: steel position (top or bottom of beam as cast), steel deformation type (old-style or A-305), stress (20,000, 30,000, 40,000, and 50,000 psi), and year (1957-1979); and three quantitative variables: condition rating, percent velocity squared (V^2), and maximum crack width. The original plan of study called for four repeated measures on the three quantitative variables for each treatment combination, i.e., position, type, stress, and year.

18. The raw data of this study were coded and entered onto the IBM 4331 computer located at the WES. The data had on-line availability for subsequent analyses using the SAS.

19. The analysis approach to this set of analyses is as follows: averages* of condition rating, %V², and maximum crack width per

* Averages were used because the SAS program cannot perform the analysis of variance procedure on interaction effects if an imbalance of replicates exists, or if there are missing replicate values. Both of these conditions exist in these data.

treatment combination. Correlation analysis by position, type of steel, and stress for condition, percent V^2 , and maximum crack width, and a four-factor (position, type, stress, and year) analysis of variance for each of the three variables with subsequent mean separations using Duncan's Multiple Range Test for significant main effects, and either John Tukey's or orthogonal mean contrasts for significant interaction effects.

20. The assumptions made for this analysis procedure are:

- a. The errors are normally distributed with a population mean of zero and an unknown variance of σ^2 .
- b. The effects of the model are fixed.

The assumption pertaining to the normal distribution may be invalid; however, the analysis of variance procedure is robust with respect to this assumption as long as the within-treatment variances are homogeneous.*

21. In order to interpret the meaning of the significant differences, an in-depth multiple comparison of the pertinent treatment combination averages was performed. For the significant main effects the Duncan's Multiple Range Test was used, and for the significant interaction effects either John Turkey's or orthogonal mean contrasts were used. The selection of the latter two as the mean separation test of choice will be discussed during the interpretation of the germane interaction effect. For an in-depth discussion of these multiple comparison procedures, reference Principles and Procedures of Statistics* by Robert G. D. Steel and James H. Torrie or Statistical Methods by George W. Snedecor and William G. Cochran.**

Variable condition

22. The analysis of variance (reference Appendix B) for the variable condition indicates that the effects of position, reinforcement bar deformation, position by reinforcement bar deformation interaction, stress, position by stress interaction, reinforcement bar deformation by stress interaction, position by reinforcement bar deformation by stress interaction, year, stress by year interaction, position by reinforcement

* R. G. D. Steel, and J. H. Torrie. 1980. Principals and Procedures of Statistics: A Biometrical Approach, 2nd ed., McGraw-Hill.

** G. W. Snedecor and W. G. Cochran. 1979. Statistical Methods, 6th ed., Iowa State University Press.

bar deformation by year interaction, and position by stress by year interaction are significant at the 0.05 level of significance.

23. For the second-order interaction effect of position by stress by year, it appears that a linear degradation trend exists for both top and bottom positions at stress levels 0 and 20,000 psi; however, for stress levels 30,000, 40,000, and 50,000 psi, departure from this linear trend exists for both the top and bottom trends (Figures 2-11).

24. For the second-order interaction effect of position by reinforcement bar deformation by year, the assumption of no departure from a linear degradation trend is not too seriously violated (Figures 12-15); however, it is apparent from these figures that the independence assumption, i.e., departure from parallel response relationships, is seriously violated. An in-depth characterization of these response relationships indicates that for the A 305-50T, the top position degrades at a faster rate than the bottom. The same trend is also noticeable for the old-style reinforcement bar deformation.

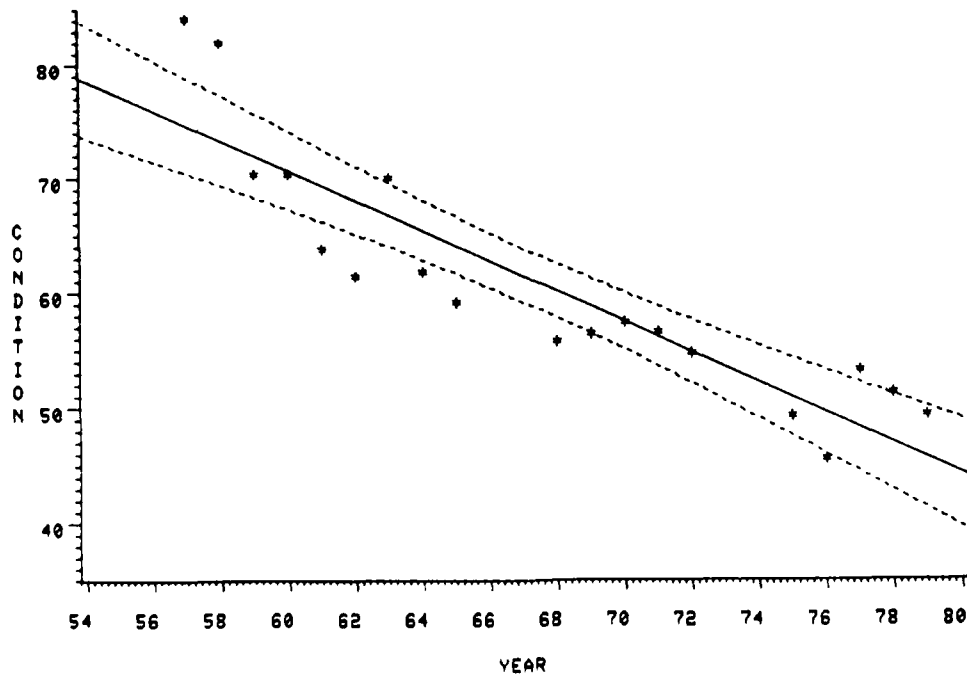


Figure 2. Condition average over reinforcement types.
Position, bottom; stress 0 psi

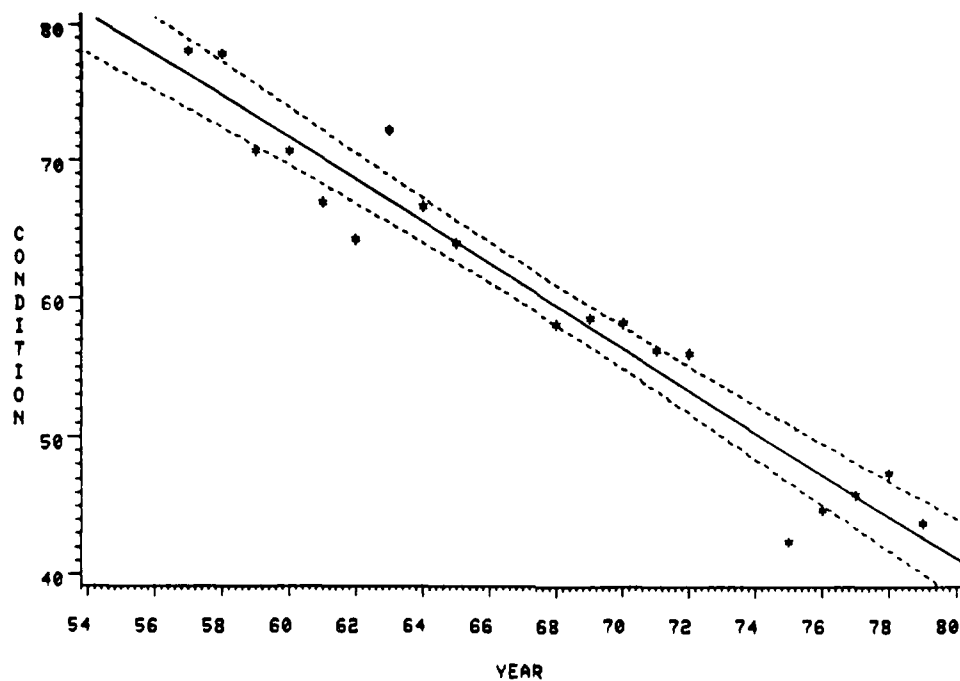


Figure 3. Condition average over reinforcement types.
Position, top; stress, 0 psi

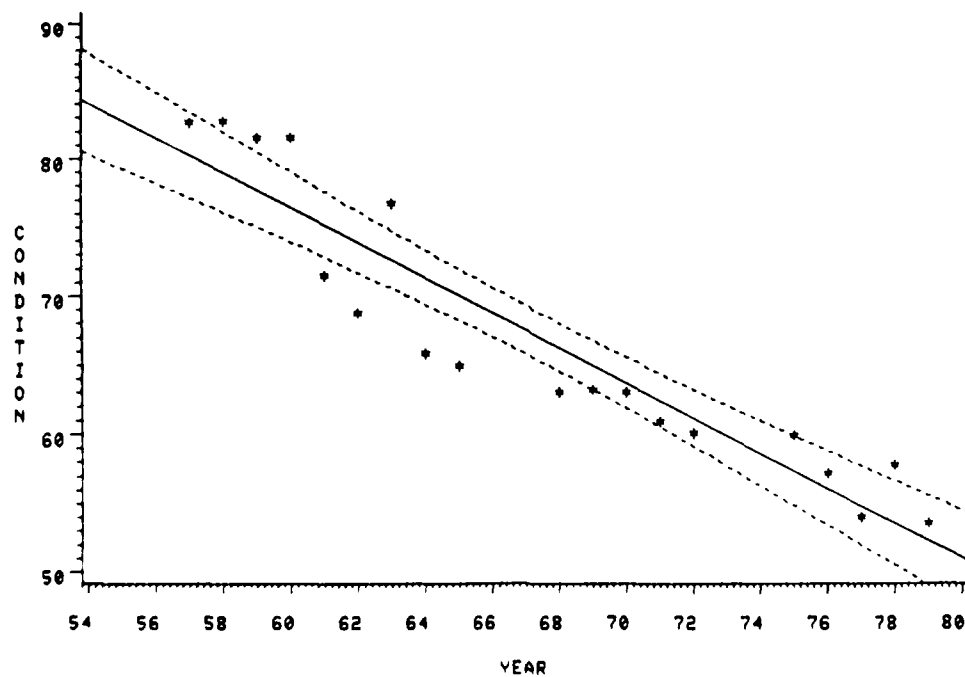


Figure 4. Condition average over reinforcement types.
Position, bottom; stress, 20,000 psi

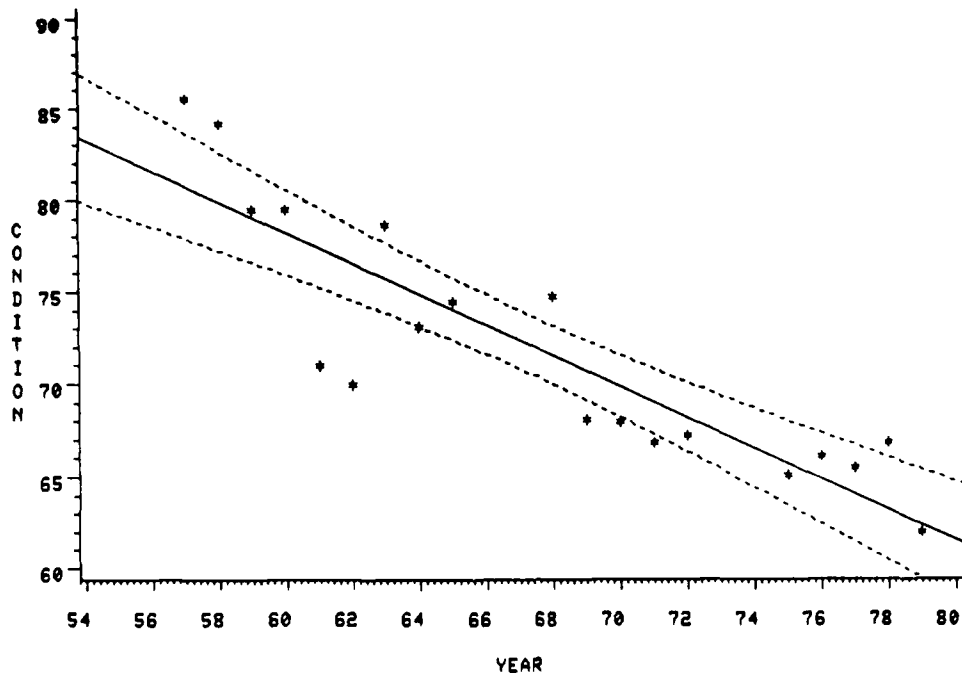


Figure 5. Condition average over reinforcement types.
Position, top; stress, 20,000 psi

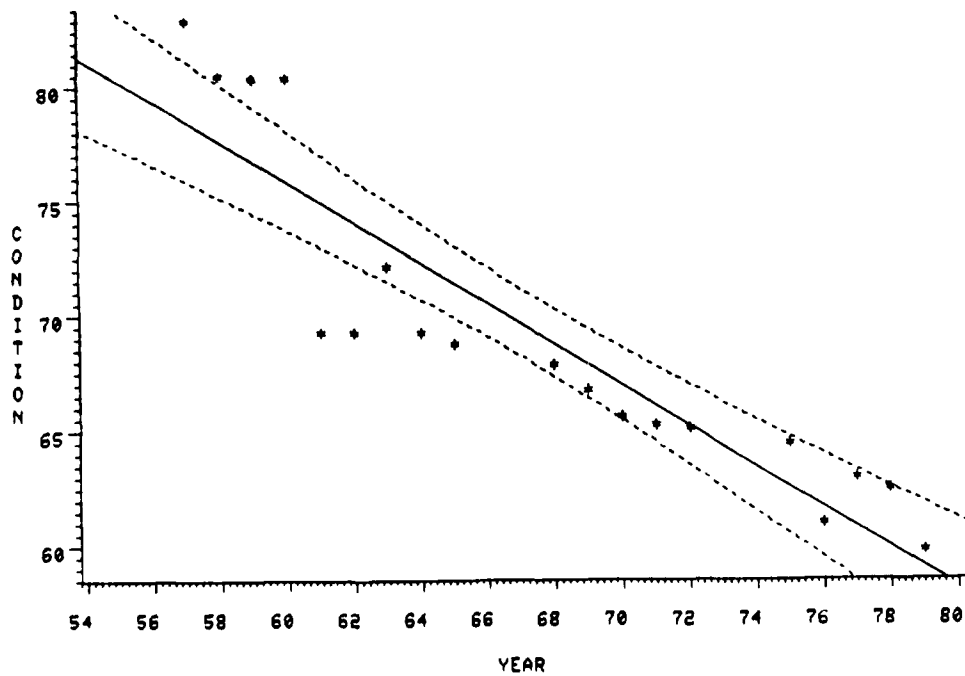


Figure 6. Condition average over reinforcement types.
Position, bottom; stress, 30,000 psi

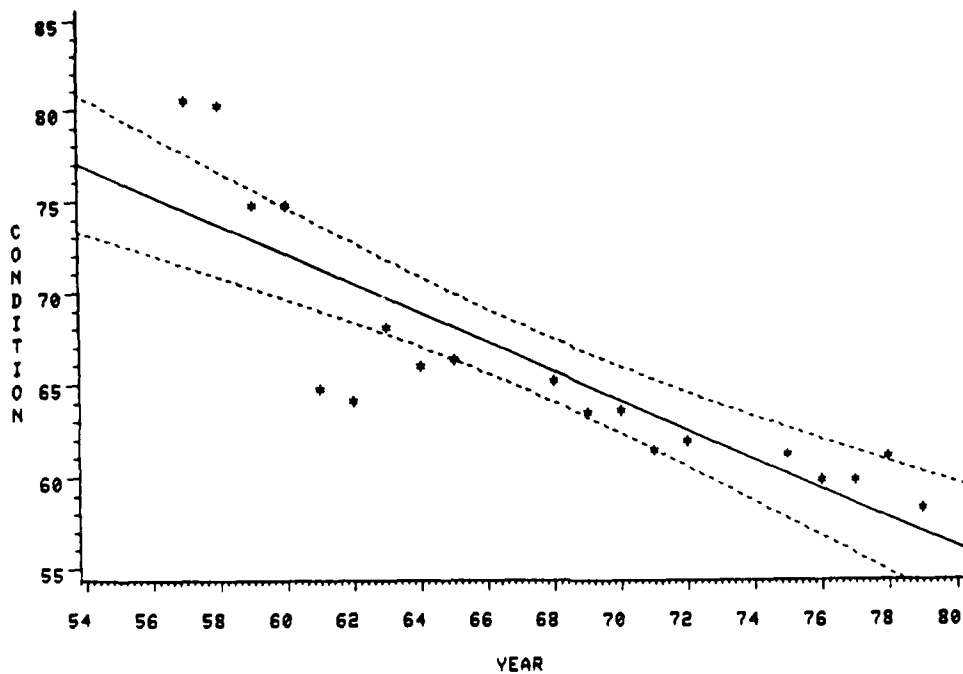


Figure 7. Condition average over reinforcement types.
Position, top; stress, 30,000 psi

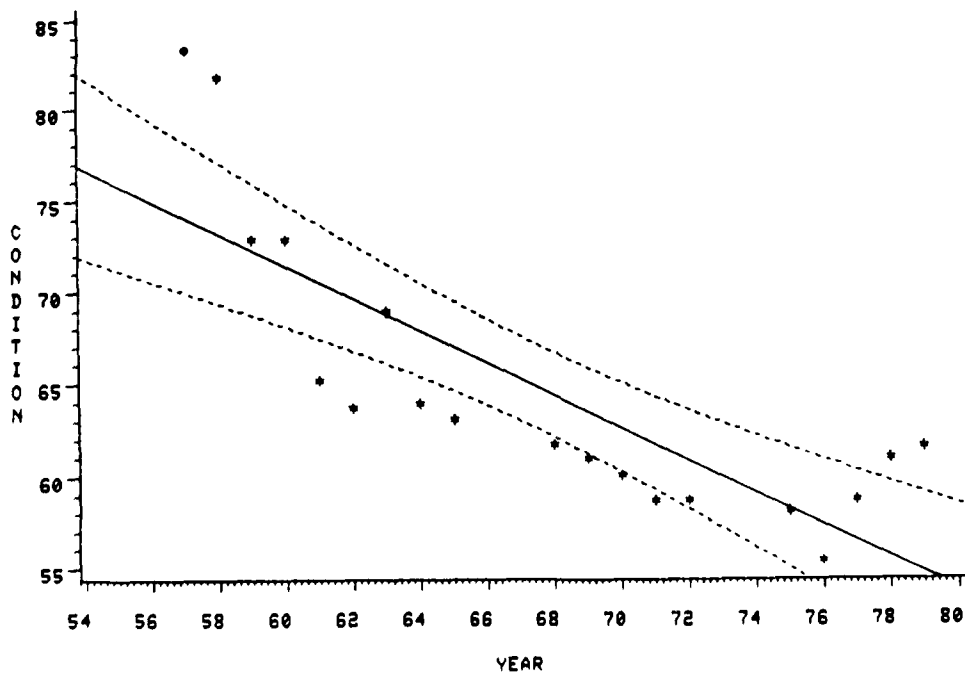


Figure 8. Condition average over reinforcement types.
Position, bottom; stress, 40,000 psi

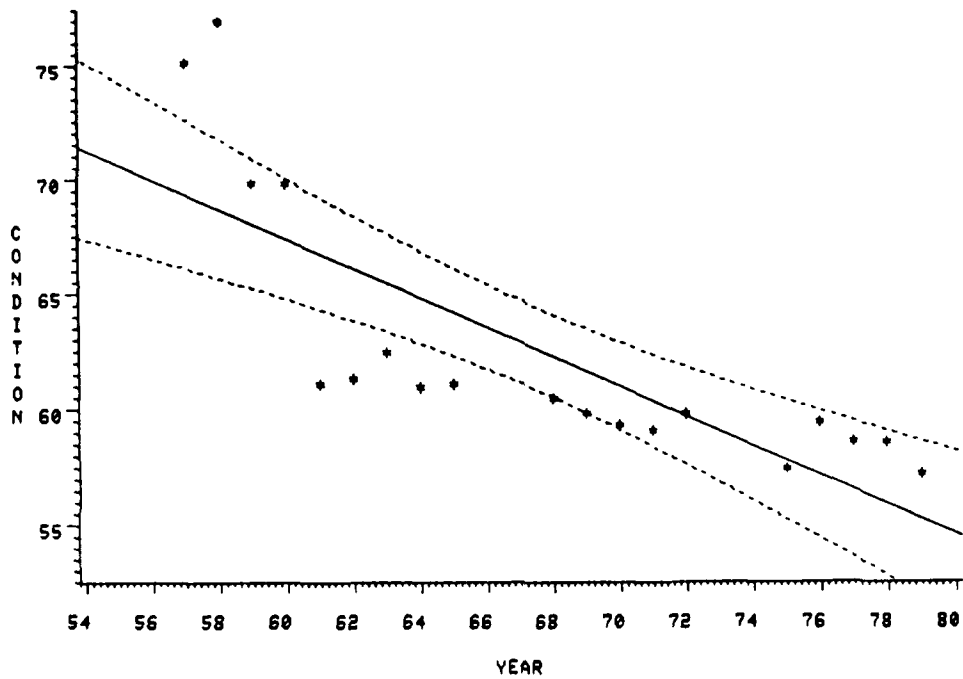


Figure 9. Condition average over reinforcement types.
Position, top; stress, 40,000 psi

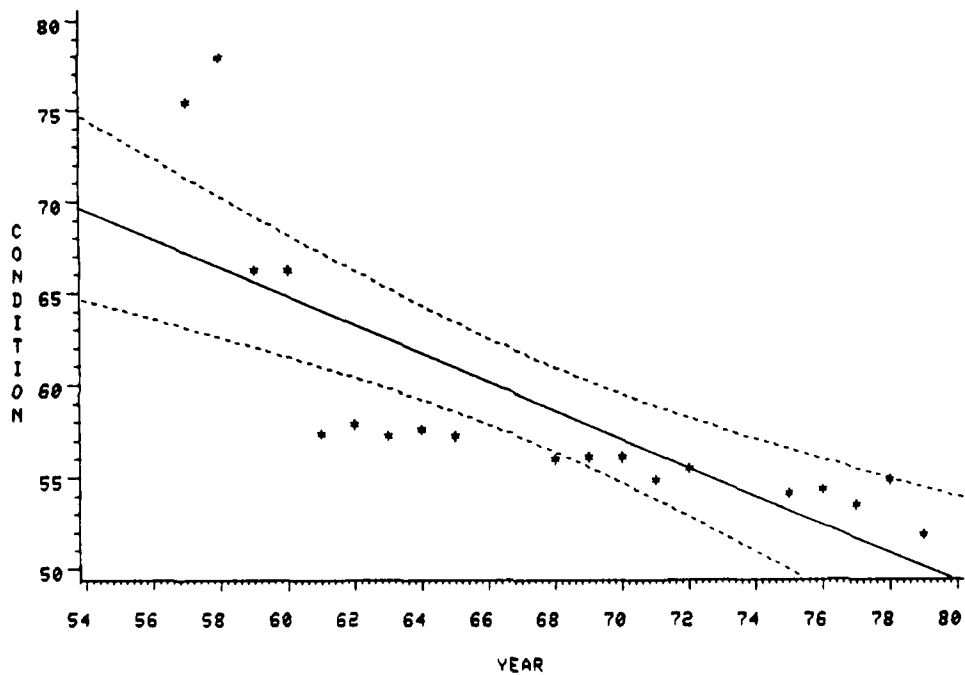


Figure 10. Condition average over reinforcement types.
Position, bottom; stress, 50,000 psi

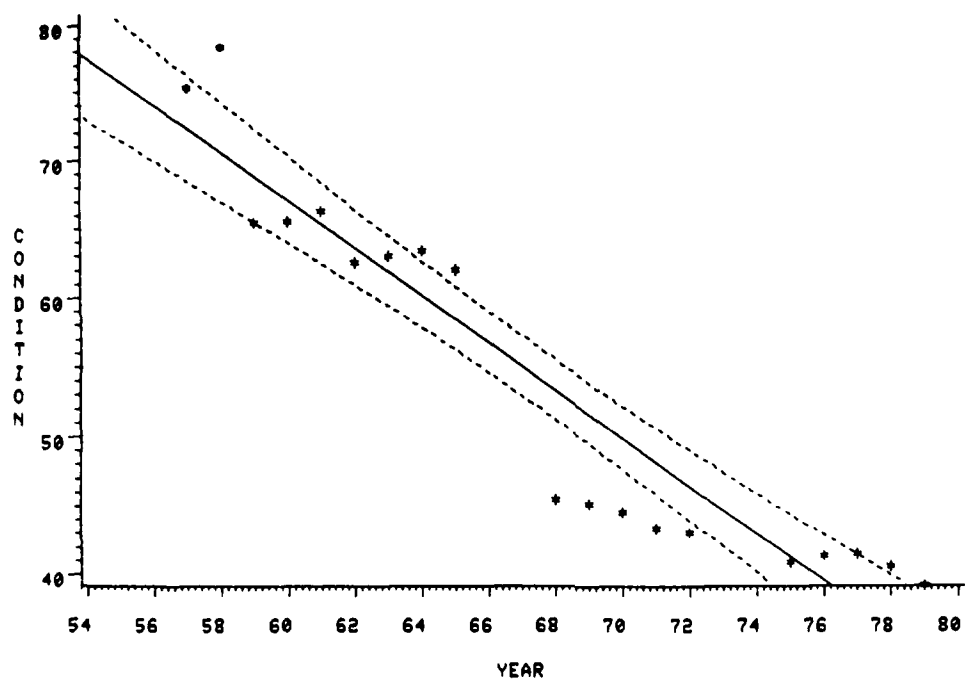


Figure 11. Condition average over reinforcement types.
Position, top; stress, 50,000 psi

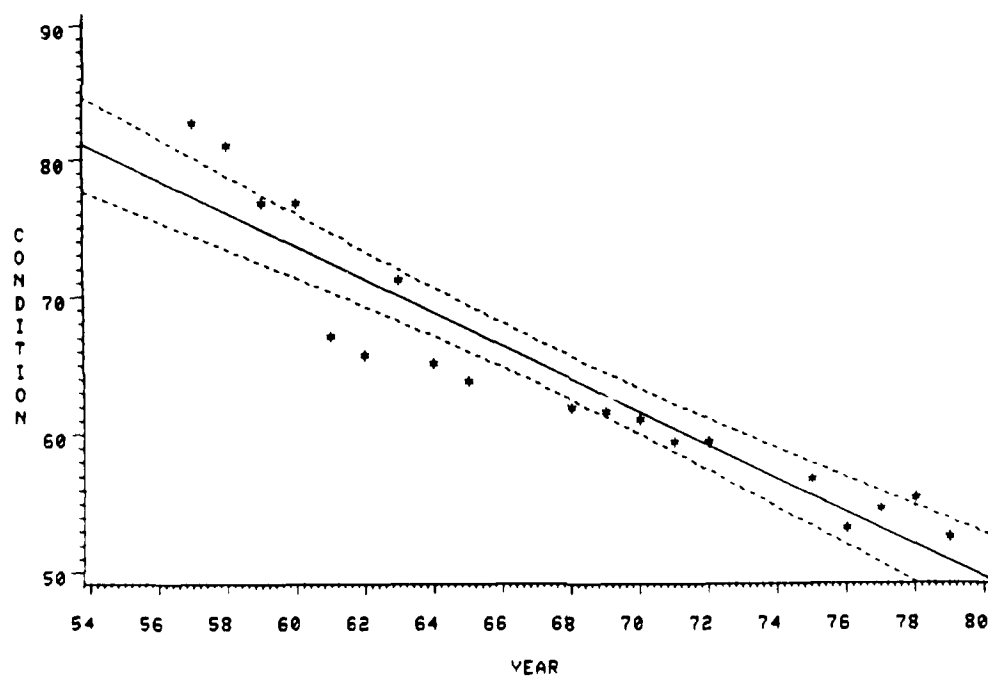


Figure 12. Condition average over stress levels.
Position, bottom; type, A-305

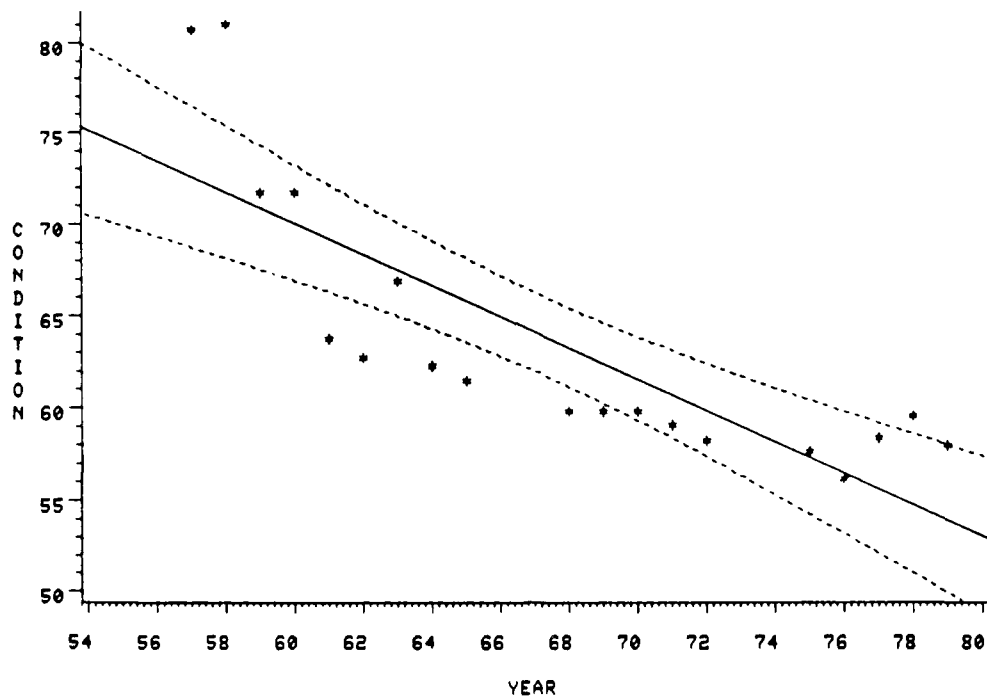


Figure 13. Condition average over stress levels.
Position, bottom; type, old-style

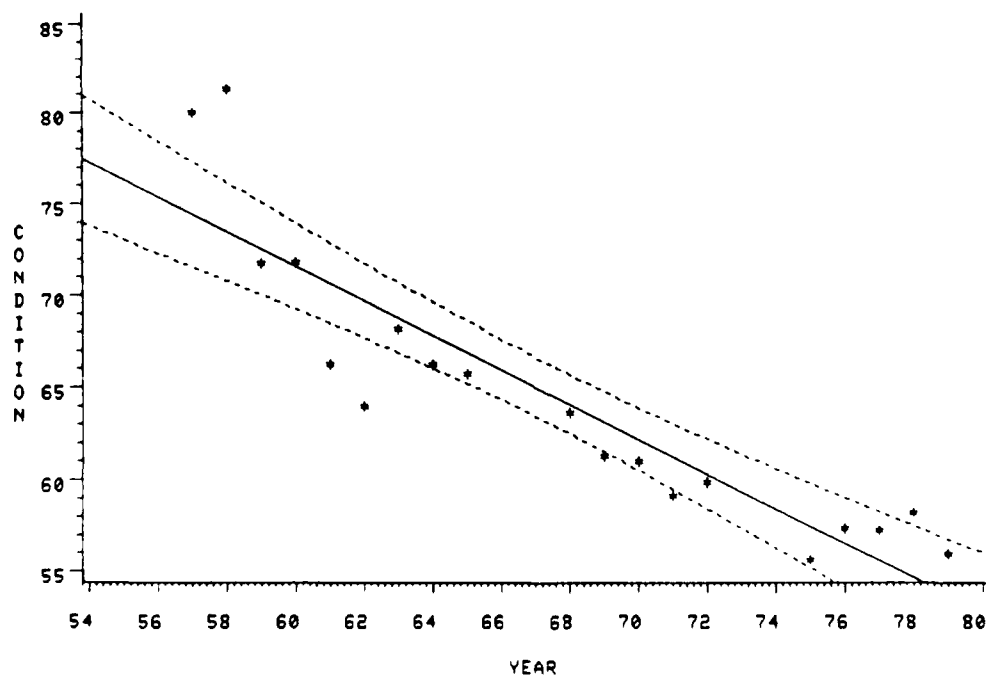


Figure 14. Condition average over stress levels.
Position, top; type, A-305

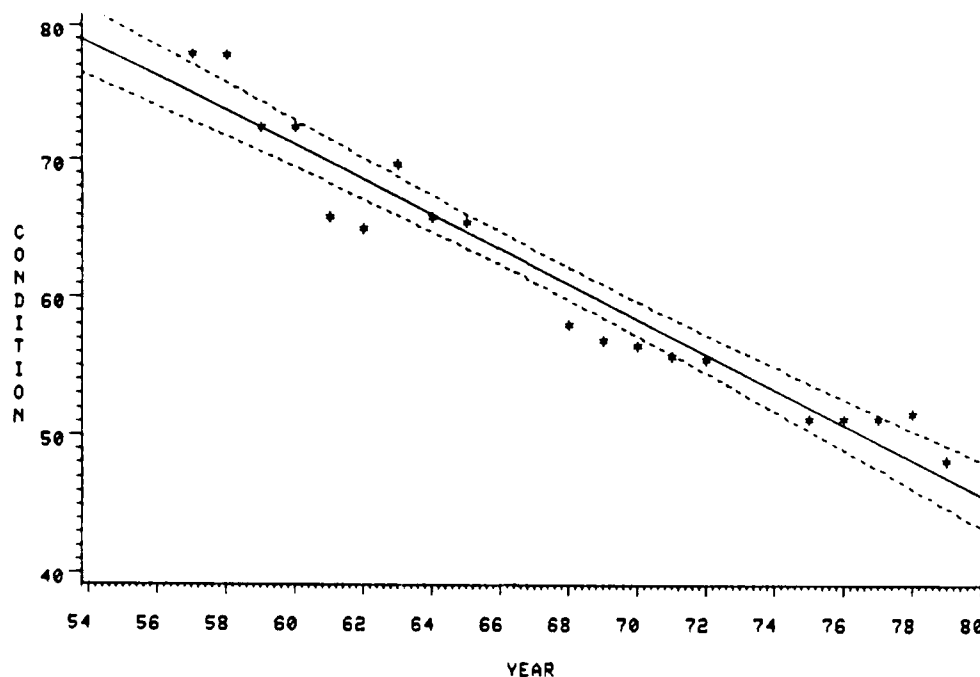


Figure 15. Condition average over stress levels.
Position, top; type, old-style

25. For the first-order interaction effect of stress by year, Table 1 displays the pertinent cell means.

Table 1
Stress by Year, Condition Ratings

| Year | Stress at | | | | |
|------|----------------|---------------|---------------|---------------|---------------|
| | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| 57 | 81 | 84 | 82 | 79 | 75 |
| 58 | 80 | 83 | 80 | 79 | 78 |
| 59 | 71 | 81 | 78 | 71 | 66 |
| 60 | 71 | 81 | 78 | 71 | 66 |
| 61 | 65 | 71 | 67 | 63 | 62 |
| 62 | 63 | 69 | 67 | 63 | 60 |
| 63 | 71 | 78 | 70 | 66 | 60 |
| 64 | 64 | 70 | 68 | 63 | 61 |
| 65 | 61 | 70 | 68 | 62 | 60 |
| 68 | 57 | 69 | 67 | 61 | 51 |

(Continued)

Table 1 (Concluded)

| Year | Stress at | | | | |
|------|----------------|---------------|---------------|---------------|---------------|
| | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| 69 | 58 | 66 | 65 | 60 | 51 |
| 70 | 58 | 66 | 64 | 60 | 50 |
| 71 | 57 | 64 | 63 | 59 | 49 |
| 72 | 56 | 64 | 64 | 59 | 49 |
| 75 | 46 | 63 | 63 | 58 | 48 |
| 76 | 45 | 62 | 60 | 57 | 48 |
| 77 | 50 | 60 | 61 | 59 | 48 |
| 78 | 49 | 62 | 62 | 60 | 48 |
| 79 | 47 | 59 | 59 | 59 | 46 |

26. The John Tukey w-procedure was used to compare the five cell means within a year category. This multiple comparison procedure uses the error mean squares from the analysis of variance table, the number of observations within each cell mean, and the upper percentage points of the studentized range which is a tabular value found in most statistical methods tests. The w-procedure utilizes the following:

$$w = Q(t, df) \sqrt{(\text{error mean squares})/N}$$

where $Q(t, df)$ is the tabular studentized range value, t is the number of means being compared, and df is the degrees of freedom of the error mean squares. It was found that the critical difference among five means composed of four observations was 7.49. Utilizing this critical difference, one may readily observe that the stress levels 20,000, 30,000, and 40,000 psi behave similarly and are significantly higher than the stress levels of 0 and 50,000 psi which behave similarly. The zero stress level (control) beams were not kept out of the sand (Figure 16) as were the yoked beams. The fact that drying could not readily occur probably accounts for the poor performance of the control beams as shown in Table 1.

27. For the second-order interaction effect of position by reinforcement bar deformation by stress, the pertinent cell means are displayed in Table 2.



Figure 16. Zero stress (control) specimens
after excavation of sand

Table 2
Position by Reinforcement Bar
Deformation by Stress, Condition Ratings

| | | Stress at | | | | |
|--------|-----------|----------------|---------------|---------------|---------------|---------------|
| | | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| Bottom | A 305-50T | 66.30 | 62.72 | 70.92 | 62.91 | 60.01 |
| | Old-Style | 55.14 | 70.93 | 67.42 | 66.70 | 57.93 |
| Top | A 305-50T | 61.05 | 73.30 | 65.24 | 58.43 | 64.30 |
| | Old-Style | 59.33 | 70.54 | 66.97 | 66.75 | 43.91 |

28. The John Tukey w-procedure was used to simultaneously compare the five cell means within position and reinforcement deformation type, and it was found that the critical difference is 3.44. This in essence means that if any two cell means within the position and deformation type differ by more than 3.44, then these two means are significantly

different. Duncan's Multiple Range notation was used to arrange these means in the following order.

| | | | | | | |
|--------|-----------|--------------|--------|--------------|--------|--------|
| | | 30,000 | 0 | 40,000 | 20,000 | 50,000 |
| Bottom | A 305-50T | <u>70.92</u> | 66.30 | <u>62.91</u> | 62.72 | 60.01 |
| | | 20,000 | 30,000 | 40,000 | 50,000 | 0 |
| | Old-Style | <u>70.93</u> | 67.42 | 66.70 | 57.93 | 55.14 |
| | | 20,000 | 30,000 | 50,000 | 0 | 40,000 |
| Top | A 305-50T | <u>73.30</u> | 65.24 | 64.30 | 61.05 | 58.43 |
| | | 20,000 | 30,000 | 40,000 | 0 | 50,000 |
| | Old-Style | <u>70.54</u> | 66.97 | 66.75 | 59.33 | 43.91 |

Means underscored by the same line are not statistically different; the permutation or arrangement of the cell means in Table 2 exhibits consistent patterns. Resultant from this, the interaction of position by reinforcement type became significant when considered by stress level.

29. Again, the poor exposure condition, i.e., partially covered with sand so that drying could not readily occur, is thought to account for the poor showing of the zero stress level beams.

30. For the first-order interaction effect of reinforcement bar deformation type by stress, the pertinent cell means are displayed in Table 3 and graphically in Figure 17.

Table 3
Reinforcement Bar Deformations
by Stress Levels, Condition Ratings

| | Stress at | | | | |
|-----------|-----------|--------|--------|--------|--------|
| | 0 | 20,000 | 30,000 | 40,000 | 50,000 |
| | (Control) | psi | psi | psi | psi |
| A 305-50T | 63.68 | 68.01 | 68.08 | 60.67 | 62.16 |
| Old-style | 57.24 | 70.74 | 67.20 | 66.72 | 50.92 |

31. John Tukey's w-procedure was used to compare the five cell means within deformation type, and it was found that the critical

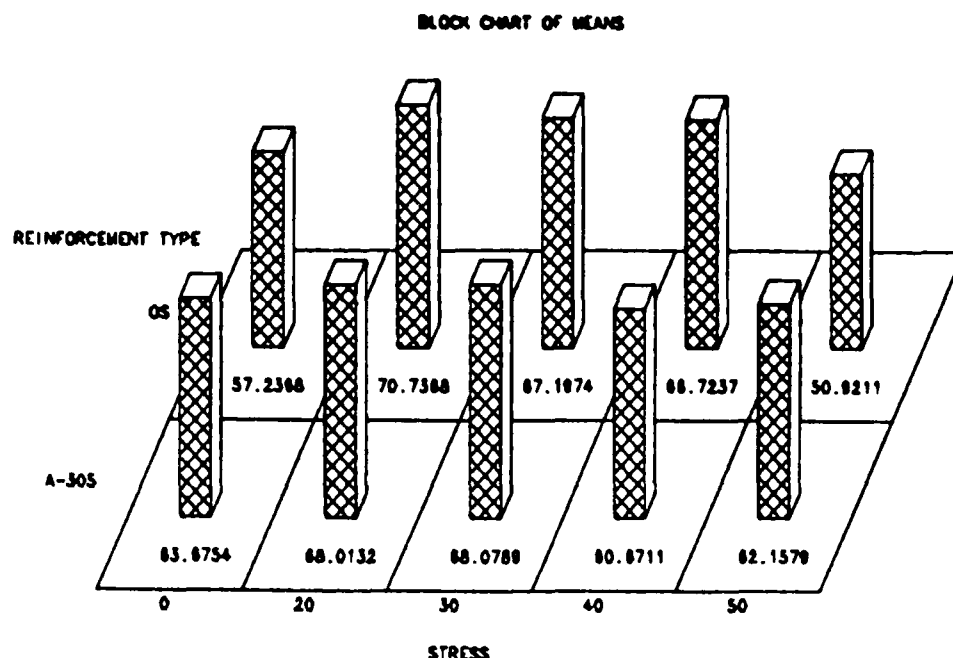


Figure 17. Condition average over years. Reinforcement types, old-style and A-305

difference is 2.28. Therefore, for the A 305-50T reinforcement bar deformation type, it is readily seen that the condition ratings of the concrete beams show a significant increase from the 0 stress level to the 20,000-psi stress level which is similar to the 30,000-psi stress level, and then the condition decreases from the 20,000- and 30,000-psi stress levels to the 40,000- and 50,000-psi stress levels, which are also similar. The old-style reinforcement bar deformation exhibited a similar pattern. Within this reinforcement deformation type, the condition of the concrete beams increased from the 0 stress level to the 20,000-psi stress level, then decreased from the 20,000- to the 30,000- and 40,000-psi stress levels, which were similar, and then exhibited a marked decrease for the 50,000-psi stress level. It is also worth noting that for the A 305-50T reinforcement type, the 0 stress level is similar to the 50,000-psi stress level; whereas with the old-style reinforcement deformation type, the 50,000-psi stress level was significantly smaller than the 0 stress level. In fact, it displayed an 11.04 percent decrease.

32. For the first-order interaction effect of position by stress, the pertinent cell means are displayed in Table 4 and graphically in Figure 18.

Table 4
Position by Stress, Condition Ratings

| | Stress at | | | | |
|--------|----------------|---------------|---------------|---------------|---------------|
| | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| Bottom | 60.72 | 66.83 | 69.17 | 64.80 | 58.97 |
| Top | 60.19 | 71.92 | 66.11 | 62.59 | 54.11 |

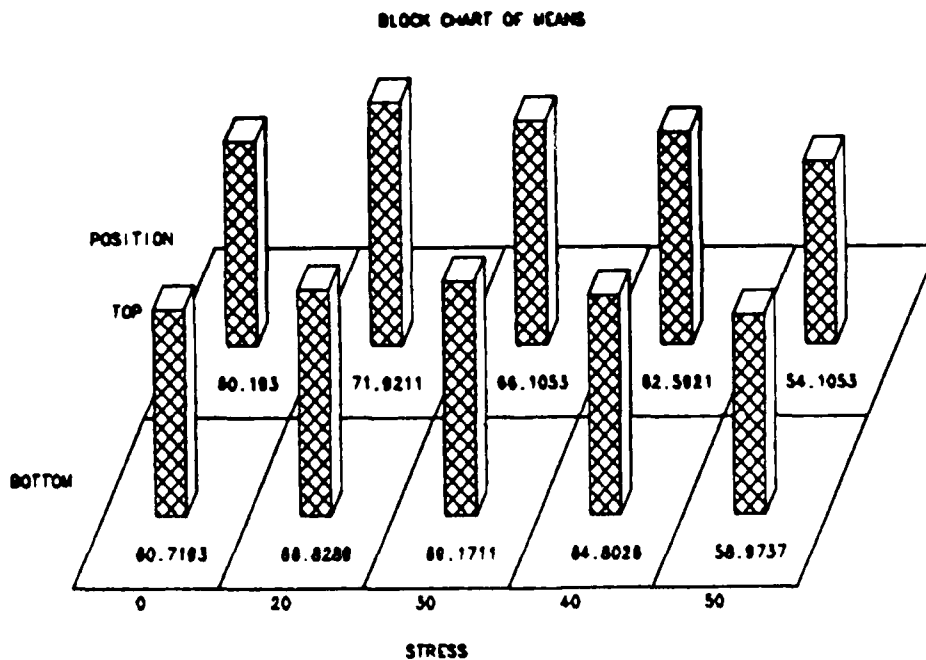


Figure 18. Condition average over years.
Positions, top and bottom

33. The John Tukey w-procedure calculation, as in the interpretation of the reinforcement bar deformation type by stress interaction, yields a critical difference of 2.28. In Table 4 similar patterns are exhibited over stress levels within the bottom position; the condition

ratings of the concrete beams increase from the 0- through the 30,000-psi stress level and then decreases from the 30,000-psi stress level through the 50,000-psi stress level, with 0- and 50,000-psi stress levels exhibiting similar condition measures. Within the top position, the condition ratings increase from 0 to 20,000 psi and then decrease through the 50,000-psi stress level, with the 50,000-psi stress level exhibiting a significant 10.10 percent decrease from the 0 stress level.

34. For the first-order interaction effect of position by reinforcement bar deformation type, the pertinent cell means are displayed in Table 5.

Table 5
Position by Reinforcement
Bar Deformation, Condition Ratings

| | <u>A 305-50T</u> | <u>Old-Style</u> | <u>Δ</u> |
|--------|------------------|------------------|----------|
| Bottom | 64.57 | 63.62 | 0.95 |
| Top | 64.47 | 61.50 | 3.20 |

35. Orthogonal contrasts were used to compare independently the cell means within position; the critical difference is 0.77. Hence, it is readily seen that the A 305-50T reinforcement bar deformation concrete beams are exhibiting significantly larger average condition rating values than the old-style reinforcement bar deformation concrete beams.

36. For the main effect of stress, Duncan's Multiple Range test produces the following pattern.

| <u>Stress at</u> | | | | |
|------------------|---------------|---------------|------------------|---------------|
| <u>20,000</u> | <u>30,000</u> | <u>40,000</u> | <u>0</u> | <u>50,000</u> |
| <u>psi</u> | <u>psi</u> | <u>psi</u> | <u>(Control)</u> | <u>psi</u> |
| <u>69.38</u> | <u>67.34</u> | <u>63.70</u> | <u>60.46</u> | <u>56.54</u> |

As can readily be seen, all stress levels are significantly different with the dominating pattern of increasing from 0 to 20,000 and decreasing from 20,000 through 50,000. The zero stress level performance is

considered to be anomolous and is probably the result of the more severe exposure conditions mentioned before, i.e., being partially covered with sand so that drying could not readily occur (Figure 16).

37. For the main effect of reinforcement bar deformation type, the A 305-50T exhibited a significantly larger average condition value (64.52) than the old-style (62.56). Also, with the main effect of position, the bottom position exhibited a significantly larger average condition rating value (64.10) than the top position (62.99).

38. Reference Appendix B for the detailed computer analysis for this data set.

39. For the response variable condition rating, the data from this investigation indicate that degradation patterns over time changed as stress levels increased. It appears that a linear degradation trend is present for the 0- and 20,000-psi stress levels; however, for the 30,000-, 40,000-, and 50,000-psi stress levels a curvilinear degradation trend is present. Also, it appears that A 305-50T reinforcement bar deformation type exhibits less severe degradation trends which do not deplete as rapidly as does the old-style reinforcement type.

Variable $\%V^2$

40. The analysis of variance for the variable $\%V^2$ indicates that the effects of reinforcement bar deformations, stress levels, position by stress level interaction, reinforcement bar deformation by stress level interaction, position by reinforcement bar deformation by stress level interaction, year, and stress by year interaction are significant at the 0.05 level of significance.

41. For the first-order interaction effect of stress by year, the pertinent cell means are displayed in Table 6. All stress levels displayed a linear degradation trend through 1972; however, an increase occurred from 1972 through 1977. Since this pattern was consistent across all stress levels, it was assumed to be an anomaly within all data sets, and was probably due to either operator differences or instrument changes or both. Regardless of the reason for the apparent anomalies, if a linear degradation response is assumed through time, then from the graphs

Table 6
Stress by Year, %V²
(Rounded to the nearest whole percent)

| Year | Stress at | | | | |
|------|----------------|---------------|---------------|---------------|---------------|
| | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| 57 | 100 | 100 | 100 | 100 | 100 |
| 58 | 102 | 102 | 103 | 104 | 102 |
| 59 | 96 | 97 | 98 | 100 | 95 |
| 60 | 99 | 85 | 84 | 88 | 78 |
| 61 | 105 | 101 | 96 | 103 | 96 |
| 62 | 100 | 104 | 99 | 104 | 96 |
| 63 | 73 | 67 | 67 | 69 | 69 |
| 64 | 80 | 69 | 65 | 70 | 70 |
| 65 | 61 | 58 | 51 | 52 | 51 |
| 68 | 56 | 50 | 48 | 49 | 42 |
| 69 | 25 | 23 | 22 | 23 | 18 |
| 70 | 39 | 40 | 36 | 34 | 30 |
| 71 | 35 | 35 | 34 | 32 | 27 |
| 72 | 30 | 26 | 27 | 27 | 20 |
| 75 | 60 | 42 | 42 | 37 | 24 |
| 76 | 61 | 46 | 46 | 40 | 34 |
| 77 | 59 | 41 | 44 | 40 | 29 |
| 78 | 35 | 30 | 35 | 32 | 21 |
| 79 | 50 | 48 | 46 | 45 | 34 |

depicted in Figures 19-28, one readily sees that the least squares regression equation shows a more rapid degradation process the higher the stress level. It is this departure from parallelism which is generating the significant stress by year interaction effect.

42. For the second-order interaction effect of position by reinforcement bar deformation by stress, the pertinent cell means are displayed in Table 7.

43. To interpret the cell means, John Tukey's w-procedure was used within each position and deformation type so that comparisons across stress levels could be performed.

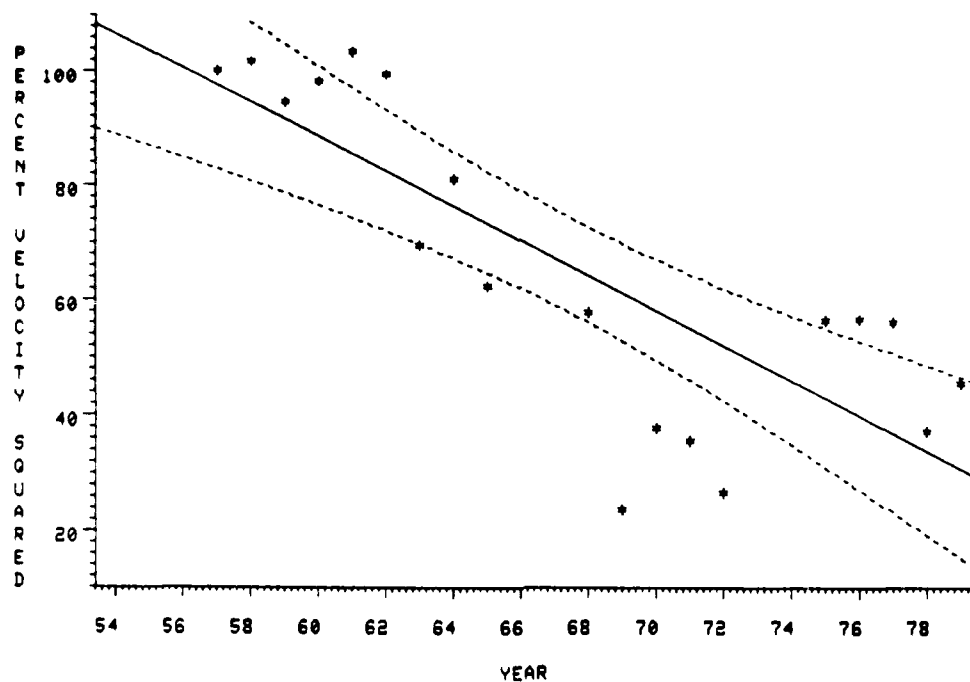


Figure 19. Percent V^2 average over reinforcement types. Position, bottom; stress, 0 psi

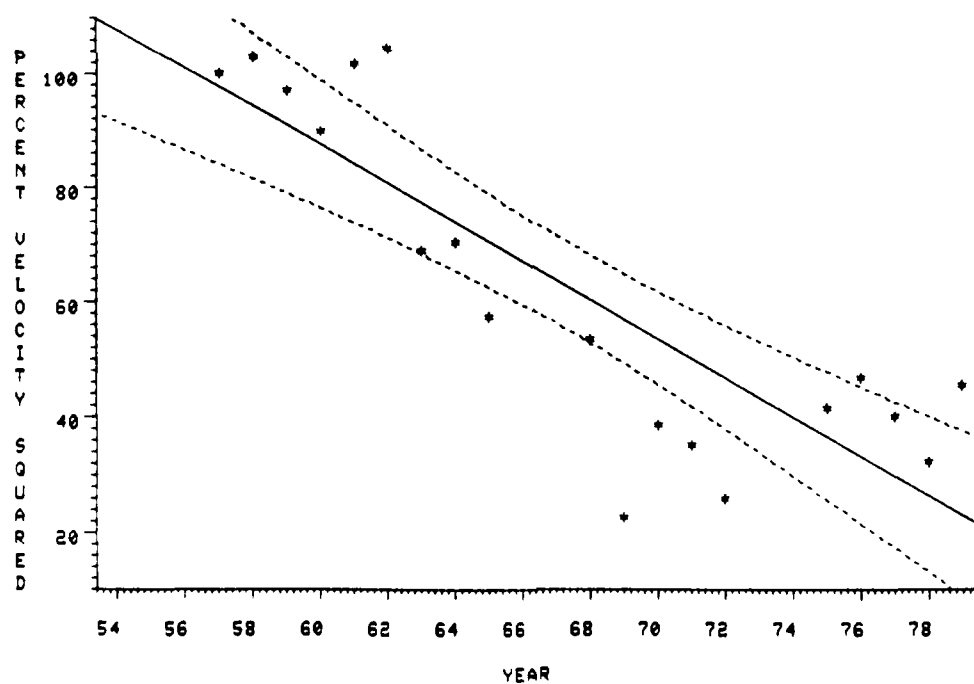


Figure 20. Percent V^2 average over reinforcement types. Position, bottom; stress, 20,000 psi

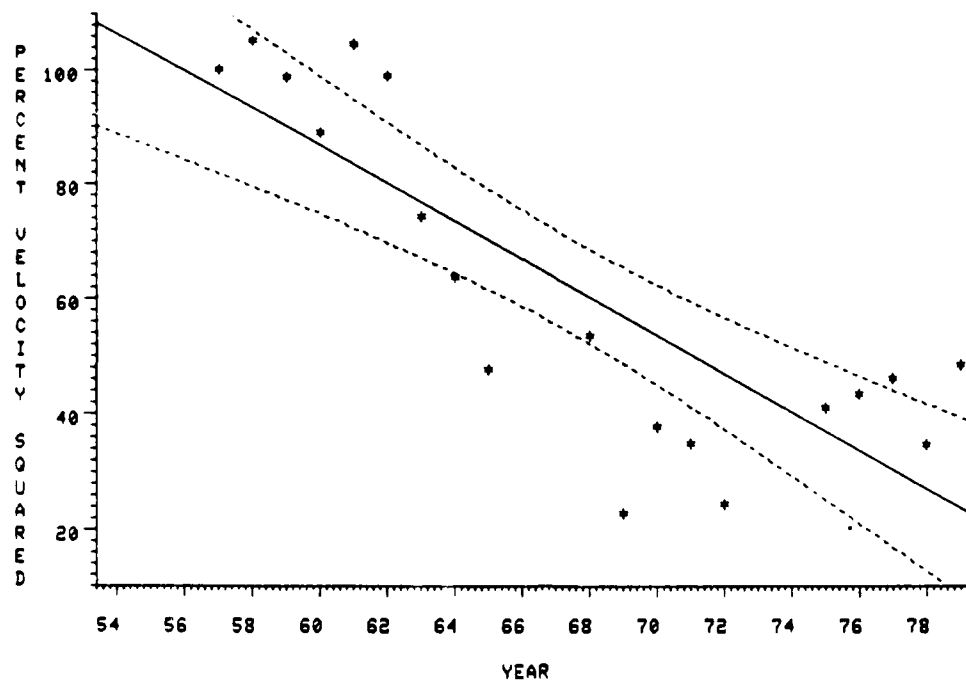


Figure 21. Percent V^2 average over reinforcement types. Position, bottom; stress, 30,000 psi

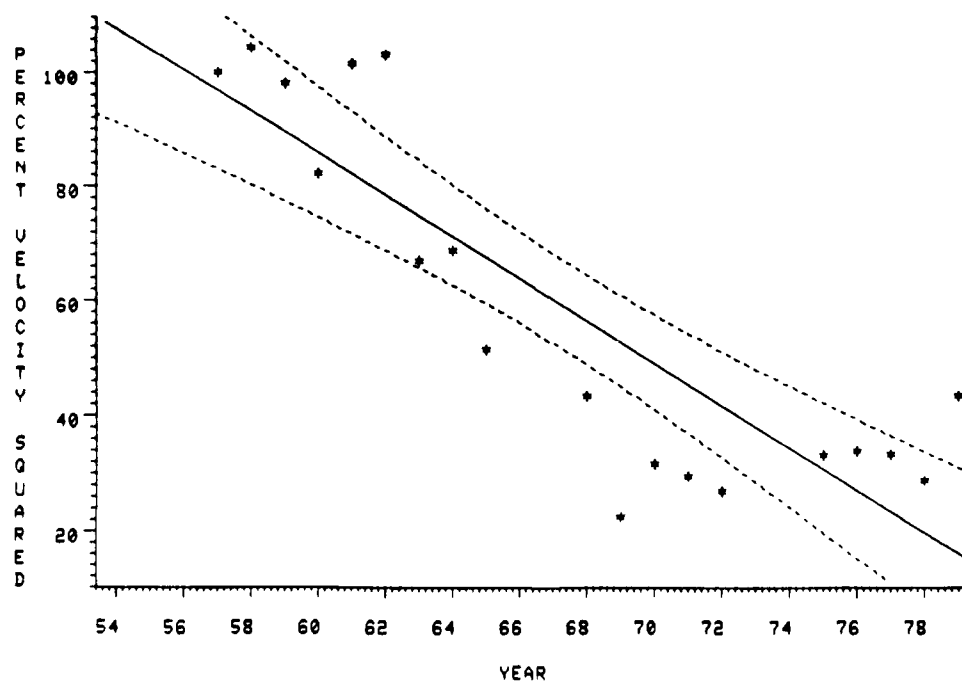


Figure 22. Percent V^2 average over reinforcement types. Position, bottom; stress, 40,000 psi

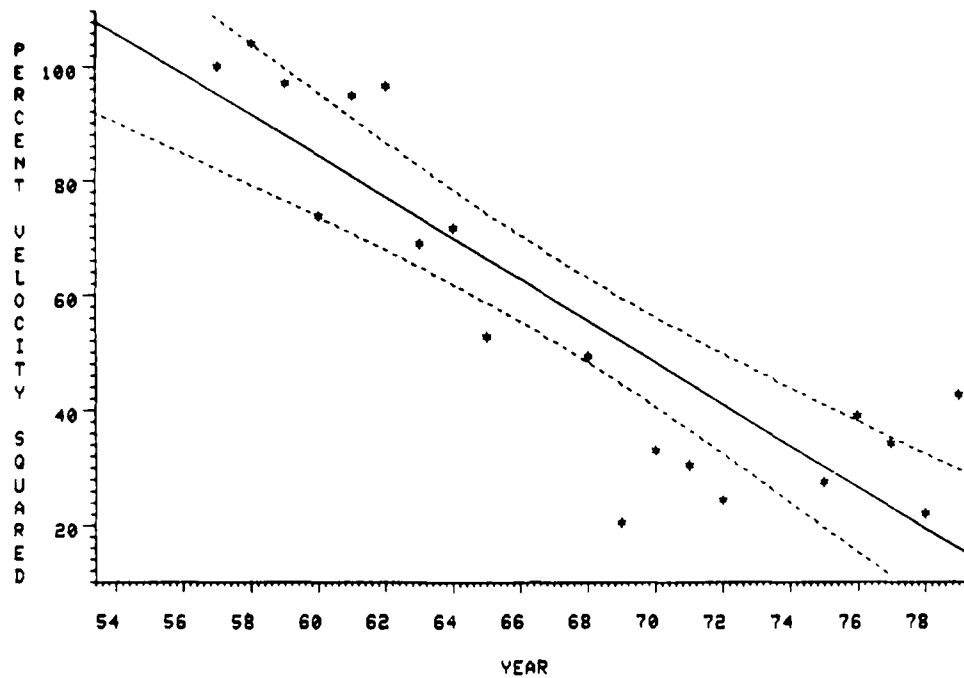


Figure 23. Percent V^2 average over reinforcement types. Position, bottom; stress, 50,000 psi

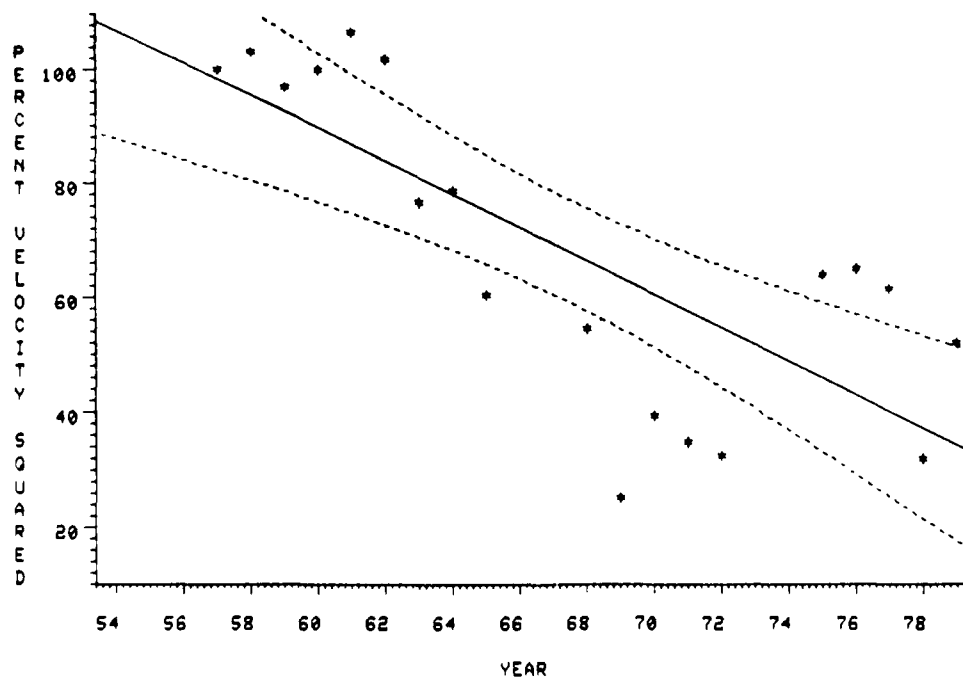


Figure 24. Percent V^2 average over reinforcement types. Position, top; stress, 0 psi

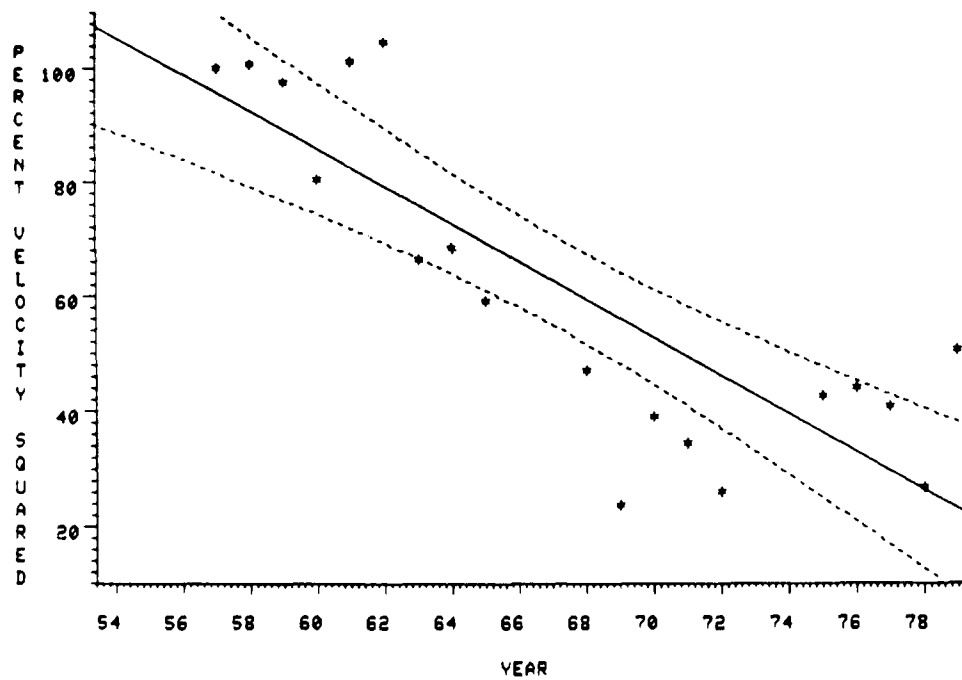


Figure 25. Percent V^2 average over reinforcement types. Position, top; stress, 20,000 psi

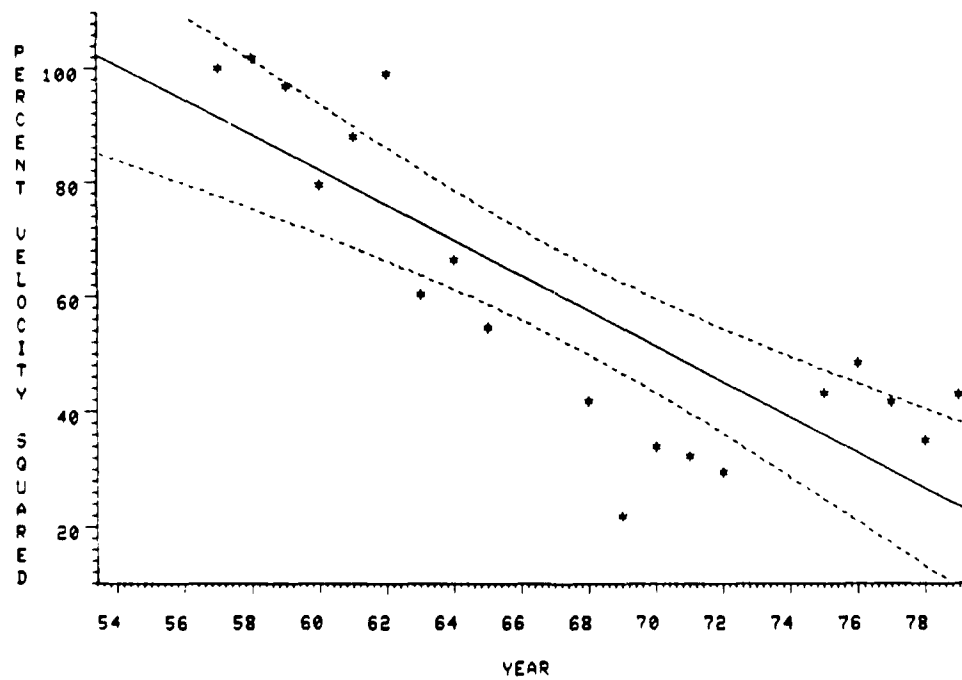


Figure 26. Percent V^2 average over reinforcement types. Position, top; stress, 30,000 psi

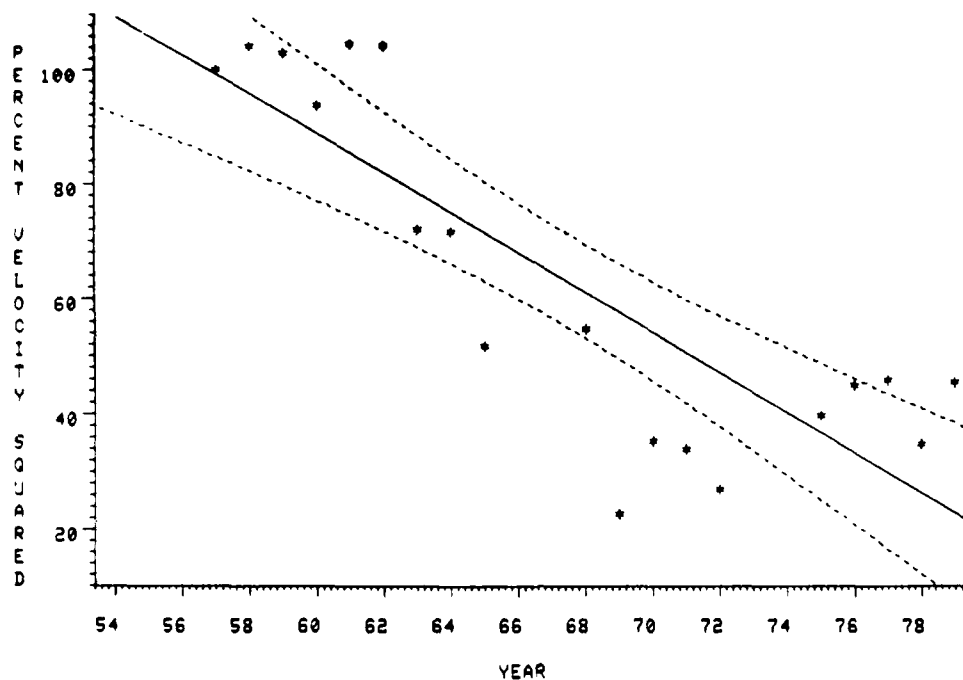


Figure 27. Percent V^2 average over reinforcement types. Position, top; stress, 40,000 psi

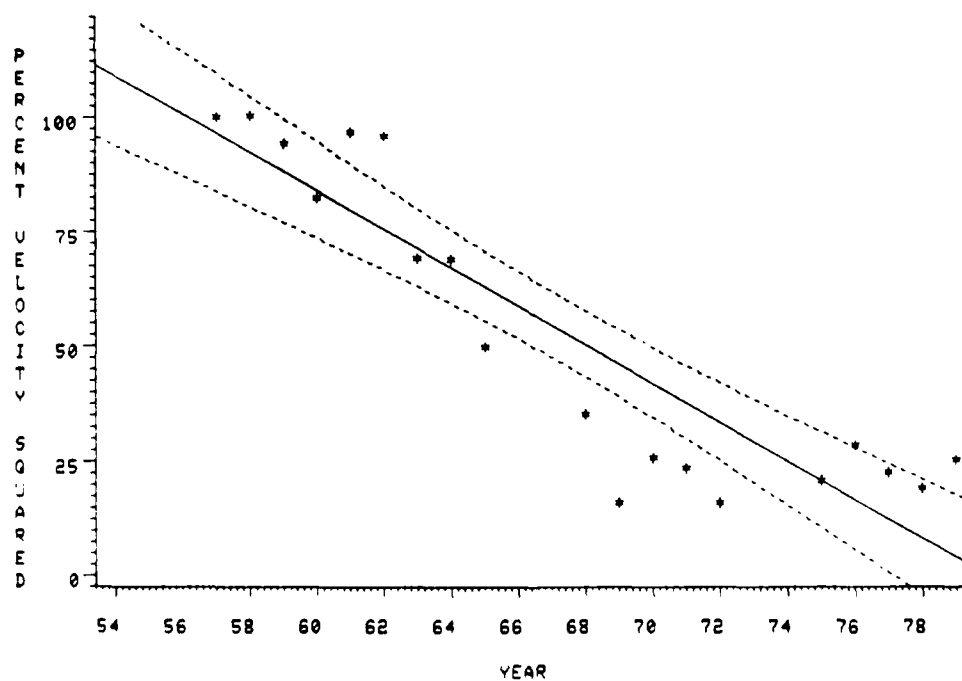


Figure 28. Percent V^2 average over reinforcement types. Position, top; stress, 50,000 psi

Table 7
Position by Deformation Type by Stress
 $\%V^2$

| | Stress at | | | | |
|-----------|----------------|---------------|---------------|---------------|---------------|
| | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| Bottom | | | | | |
| A 305-50T | 66.34 | 63.14 | 62.04 | 59.28 | 55.11 |
| Old-Style | 64.27 | 60.55 | 60.97 | 56.97 | 58.90 |
| Top | | | | | |
| A 305-50T | 67.13 | 60.18 | 59.35 | 60.73 | 58.77 |
| Old-Style | 68.16 | 61.34 | 58.26 | 64.57 | 45.49 |

For these particular cell means, the critical value of w is 4.79.
This value of w produces the following statistical patterns.

| | <u>0</u> | <u>20,000</u> | <u>30,000</u> | <u>40,000</u> | <u>50,000</u> |
|-------------------|----------|---------------|---------------|---------------|---------------|
| Bottom: A 305-50T | 66.34 | 63.14 | 62.04 | 59.28 | 55.11 |
| Bottom: Old-Style | 64.27 | 60.55 | 60.97 | 56.97 | 58.90 |
| Top: A 305-50T | 67.13 | 60.18 | 59.35 | 60.73 | 58.77 |
| Top: Old-Style | 68.16 | 61.34 | 58.26 | 64.57 | 45.99 |

In order to represent the last category, the cell means must be reordered as follows:

| <u>0</u> | <u>40,000</u> | <u>20,000</u> | <u>30,000</u> | <u>50,000</u> |
|----------|---------------|---------------|---------------|---------------|
| 68.16 | 64.57 | 61.34 | 58.26 | 45.49 |

Note: Means underscored with the same line are statistically equivalent.

44. From this type of synopsis, the changes in the significance patterns are readily seen. Since these changes are prevalent in this set of data, the interaction term became significant. From this set of cell means one would conclude the following.

- a. For the A 305-50T deformation type.
- (1) Within the bottom position, the 0 stress level has a significantly higher $\%V^2$ than the 50,000-psi stress level; furthermore, the 20,000-, 30,000-, and 40,000-psi stress levels yield similar $\%V^2$ values.
 - (2) Within the top position, the 0 stress level has a significantly larger $\%V^2$ than the 20,000-, 30,000-, 40,000-, and 50,000-psi stress levels which exhibit a similar $\%V^2$ pattern.
- b. For the old-style deformation type.
- (1) Within the bottom position, the 0, 20,000-, and 30,000-psi stress levels exhibit similar $\%V^2$ values; however, the 20,000-, 30,000-, 40,000-, and 50,000-psi stress levels also exhibit similar $\%V^2$ values. Therefore, the primary conclusion would be that 0 stress level produces a significantly larger $\%V^2$ than the 40,000- and 50,000-psi stress levels.
 - (2) Within the top position, the pattern is more complex; however, the conclusions that the 0 stress level exhibits significantly larger $\%V^2$ than the 20,000- and 30,000-psi stress levels and that the 20,000- and 30,000-psi stress levels exhibit significantly larger $\%V^2$ than the 50,000-psi stress level can be inferred.

45. For the first-order interaction effect of reinforcement bar deformation by stress, the pertinent cell means are displayed in Table 8. A three-dimensional block chart of the cell means is presented in Figure 29.

Table 8
Reinforcement Bar Deformation by Stress Levels, $\%V^2$

| Reinforcement Type | Stress at | | | | |
|-----------------------|----------------|---------------|---------------|---------------|---------------|
| | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| A 305-50T | 66.88 | 61.66 | 60.94 | 60.01 | 56.94 |
| Old-Style | 66.21 | 60.94 | 59.61 | 60.77 | 52.20 |
| Difference | 0.67 | 0.72 | 1.33 | -0.76 | 4.74 |

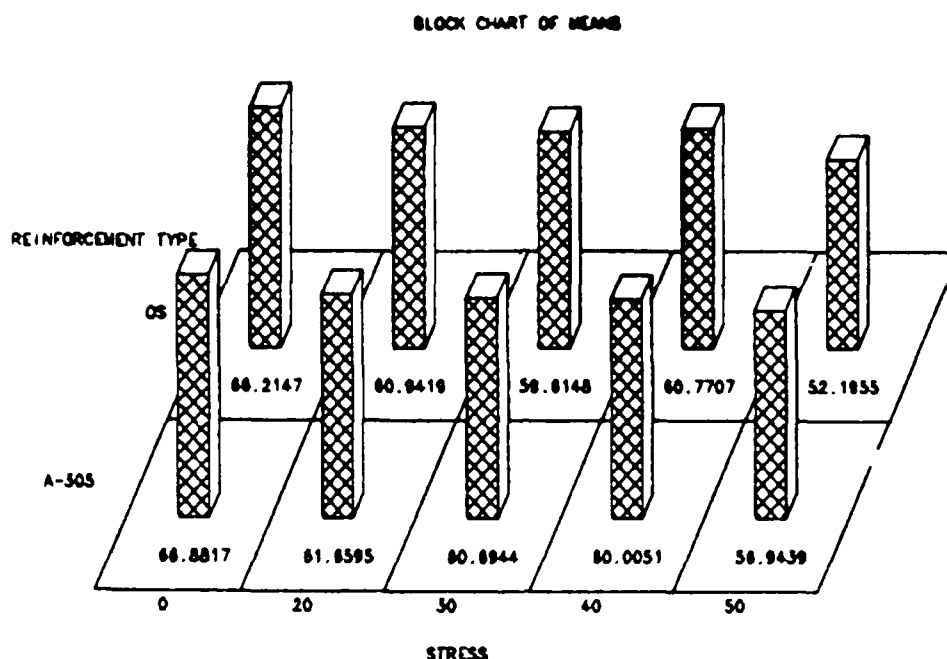


Figure 29. $\%V^2$ average over years by deformation type

The multiple comparison procedure known as orthogonal comparisons was used to determine the critical difference between any two cell means within stress levels; the critical difference is 1.81, which is obtained by

$$\text{Critical difference} = t(p, df) \sqrt{(\text{error mean squares})/N}$$

This equation is used for independent or orthogonal contrasts, where $t(p, df)$ is the tabular point from the t-distribution with degrees of freedom (df) and confidence level $(1 - p)$. N represents the number of observations comprising each cell mean. With 1.81 as the defined critical difference, the only significant difference occurs at the 50,000-psi stress level, where the A 305-50T reinforcement deformation type exhibits a significantly larger $\%V^2$ value than the old-style deformation type.

46. For the first-order interaction effect of position by stress, the pertinent cell means are displayed in Table 9 and are graphically depicted in Figure 30.

Table 9
Position by Stress, $\%V^2$

| Position | Stress at | | | | |
|------------|----------------|---------------|---------------|---------------|---------------|
| | 0 (Control) | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| Bottom | 65.45 | 61.84 | 61.51 | 58.12 | 57.01 |
| Top | 67.64 | 60.76 | 58.80 | 62.65 | 52.13 |
| Difference | -2.19 | 1.08 | 2.71 | -4.53 | 4.88 |

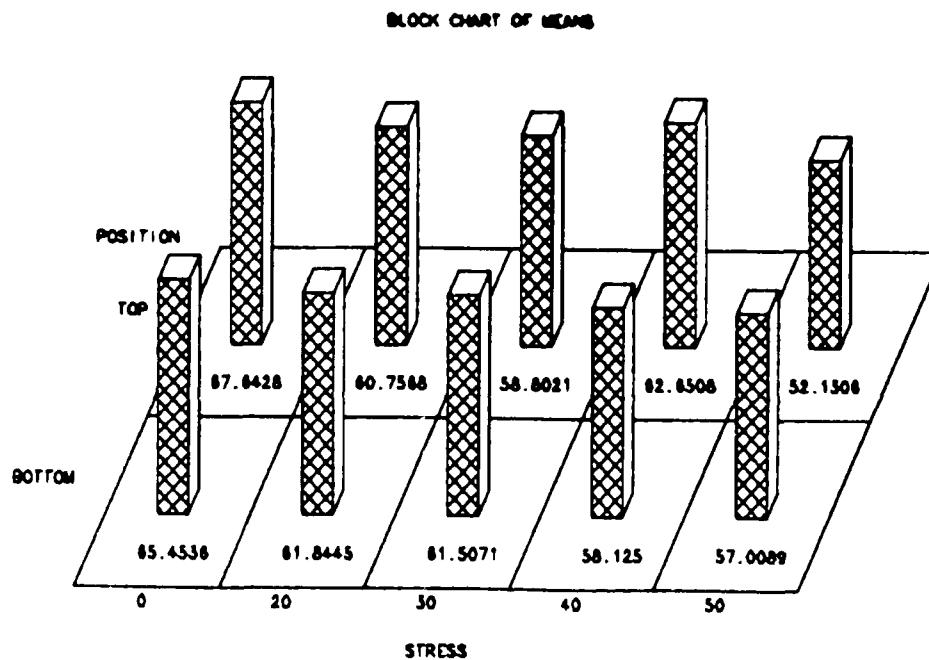


Figure 30. $\%V^2$ average over years by position

47. The absolute critical difference for this set of cell means is 1.81. As can be seen from Table 9, it can be concluded that at the 0 stress level the top position exhibited a significantly larger $\%V^2$ than the bottom position; no significant differences are exhibited at the 20,000-psi stress level; at the 30,000-psi stress level, the bottom position exhibits a significantly larger $\%V^2$ than the top position; at the 40,000-psi stress level, the top position exhibits a significantly

larger $\%V^2$ than the bottom position; and at the 50,000-psi stress level, the bottom position exhibits a significantly larger $\%V^2$ than the top position.

48. For the main effect of stress, Duncan's Multiple Range test indicates that the 0 stress level exhibits a significantly larger $\%V^2$ than the 20,000-, 30,000-, and 40,000-psi stress levels which in turn are significantly larger than the 50,000-psi stress level.

| 0 (Control) | Stress at | | | |
|----------------|---------------|---------------|---------------|---------------|
| | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| <u>66.55</u> | <u>61.30</u> | <u>60.39</u> | <u>60.15</u> | <u>54.57</u> |

Reference Appendix B for the detailed computer analysis of this data set.

49. For the main effect of reinforcement deformation type, the A 305-50T deformation type exhibits a significantly higher $\%V^2$ than the old-style.

| A 305-50T | Old-Style |
|--------------|--------------|
| <u>61.24</u> | <u>59.95</u> |

Reference Appendix B for the detailed computer analysis for this data set.

50. For the response variable $\%V^2$, the data from this investigation indicated that the degradation rate of $\%V^2$ increases as stress levels increase (exhibited by the significant stress by year interaction); the mean $\%V^2$ averaged over time indicated that the average $\%V^2$ decreased as stress increased, and that the primary difference between the A 305-50T and the old-style deformation types occurred at the 50,000-psi stress level where the A 305-50T deformation type exhibited a significantly larger $\%V^2$.

Variable
maximum crack width

51. For the variable maximum crack width, the 0 stress level was omitted due to the absence of measurable cracks. However, with stress levels of 20,000, 30,000, 40,000, and 50,000 psi, the analysis of

variance procedure indicated that the following factors were significant: stress levels, reinforcement deformation type by stress interaction, position by reinforcement deformation type by stress interaction, year, and stress by year interaction. Subsequent analyses of these significant effects are described below.

52. For the first-order interaction effect of stress by year, the data are graphically displayed in Figures 31-38. From these plots it is readily seen that maximum crack width tends to increase linearly with age for the stress levels of 20,000, 30,000, and 40,000 psi; however, for the 50,000-psi stress level there is definitely a nonlinear relationship. Maximum crack widths within the 50,000-psi stress level group display a fairly smooth linear trend until 1975, and then a more rapidly linear increasing trend through 1979.

53. For the second-order interaction effect of position by deformation type by stress level, the data are displayed in Table 10. As is readily observed from this table, maximum crack width displays a slight

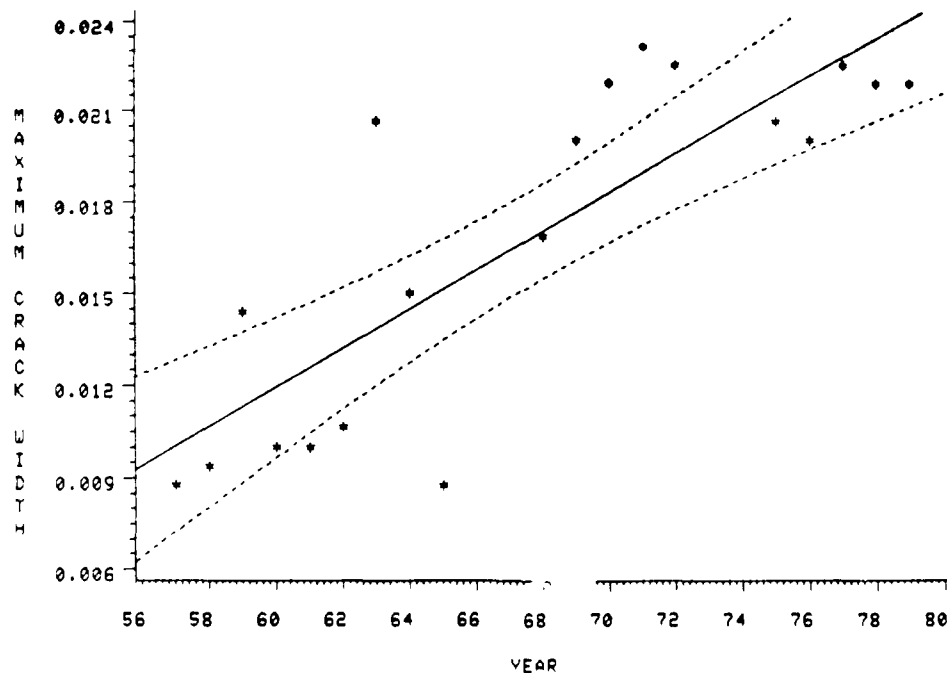


Figure 31. Maximum crack width average over reinforcement types. Position, bottom; stress, 20,000 psi

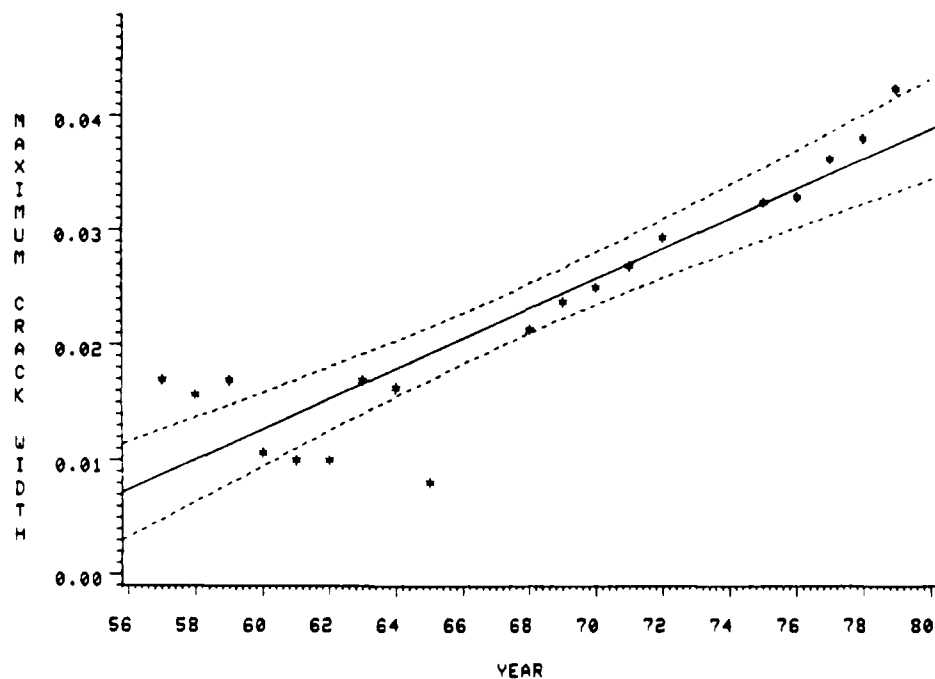


Figure 32. Maximum crack width average over reinforcement types. Position, top; stress, 20,000 psi

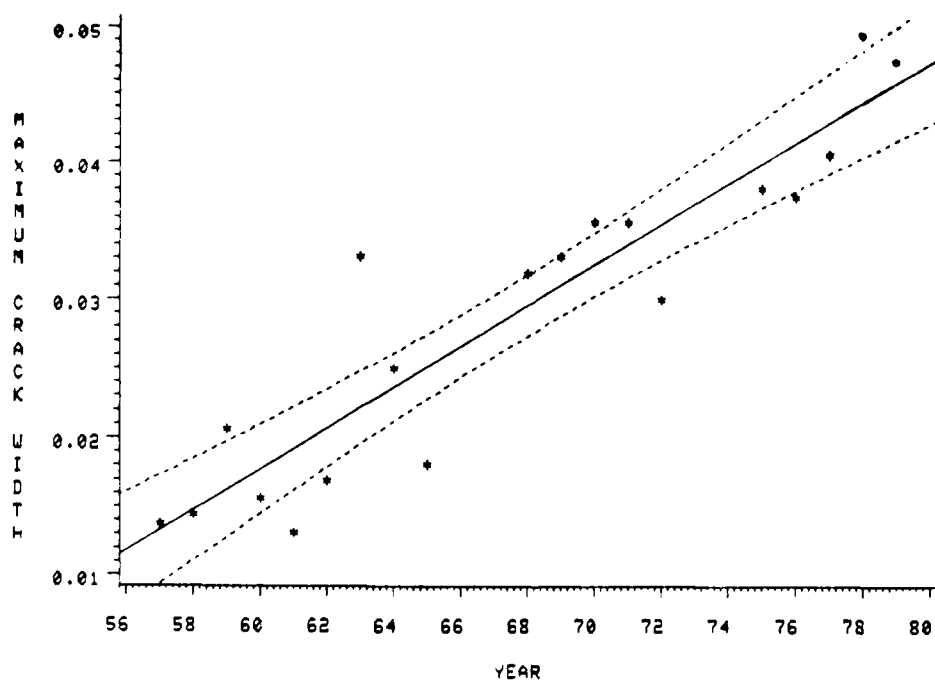


Figure 33. Maximum crack width average over reinforcement types. Position, bottom; stress, 30,000 psi

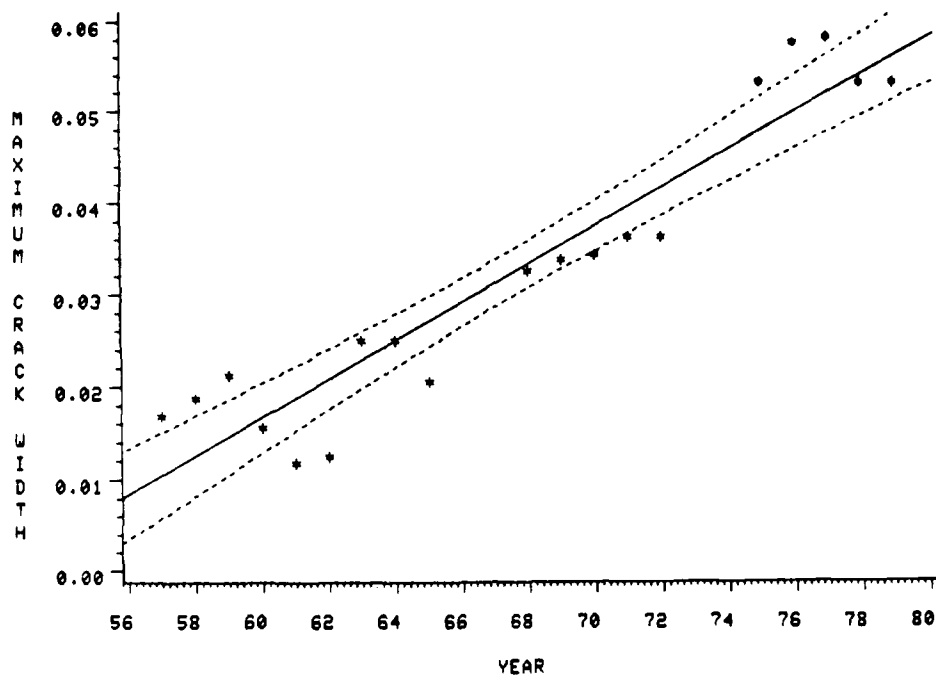


Figure 34. Maximum crack width average over reinforcement types. Position, top; stress, 30,000 psi

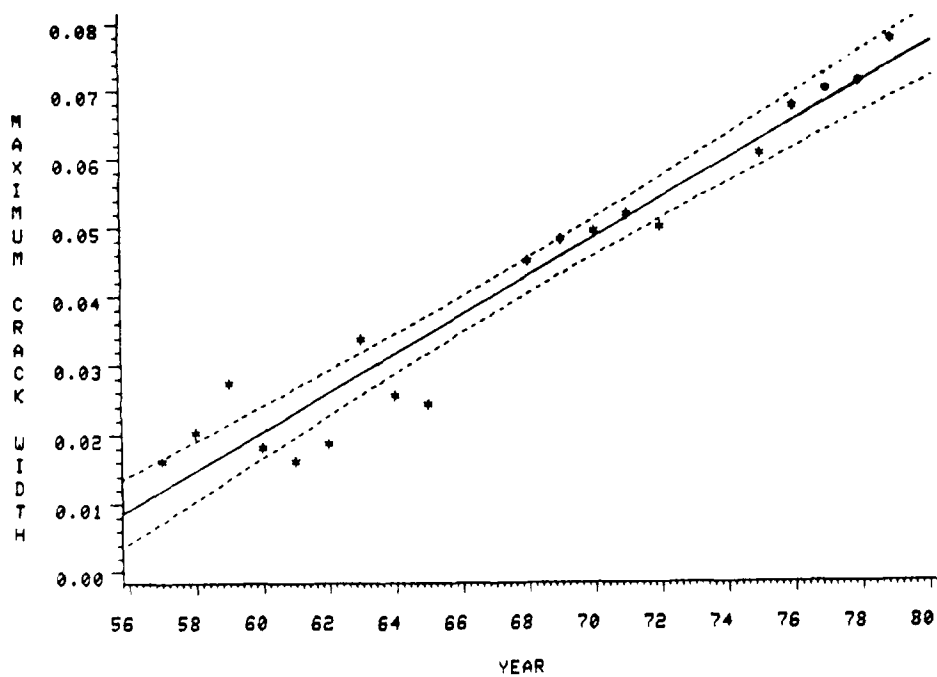


Figure 35. Maximum crack width average over reinforcement types. Position, bottom; stress, 40,000 psi

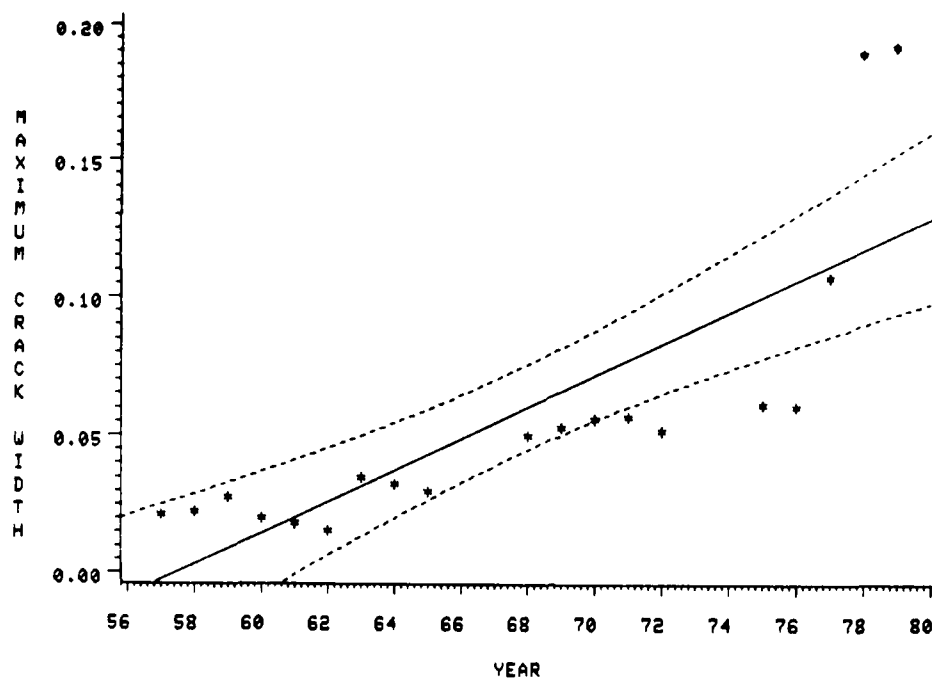


Figure 36. Maximum crack width average over reinforcement types. Position, top; stress, 40,000 psi

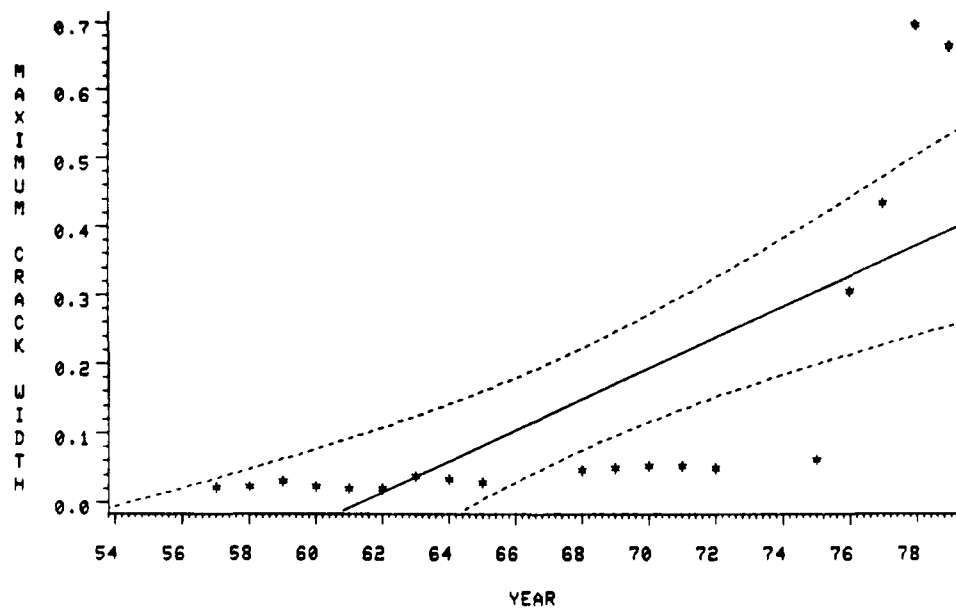


Figure 37. Maximum crack width average over reinforcement types. Position, bottom; stress, 50,000 psi

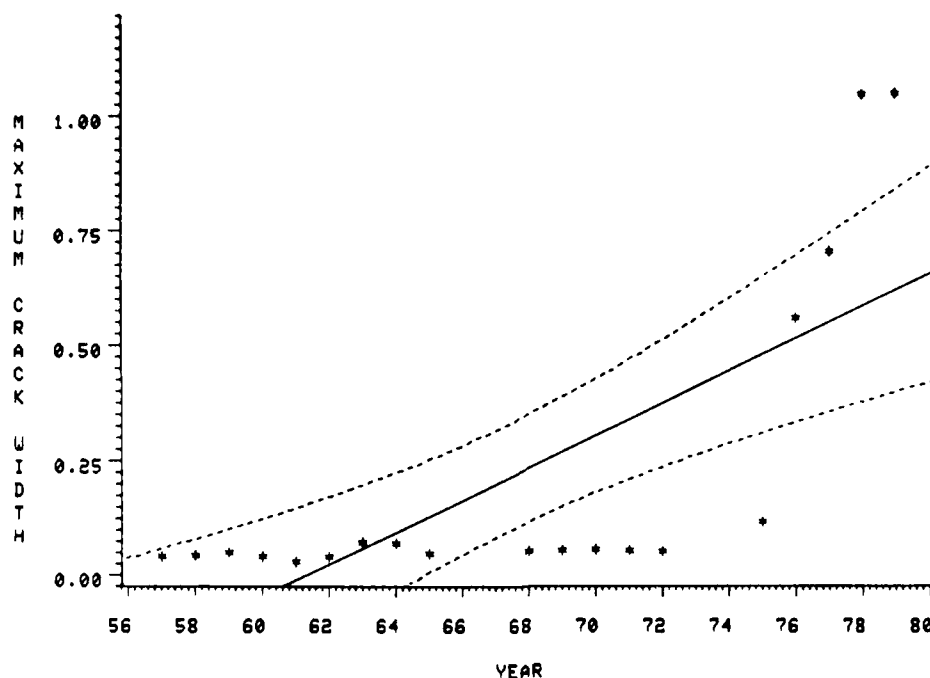


Figure 38. Maximum crack width average over reinforcement types. Position, top; stress, 50,000 psi

Table 10
Position by Stress by Reinforcement Bar Deformation,
Maximum Crack Width (in.)

| Position/Type | Stress at | | | |
|---------------|---------------|---------------|---------------|---------------|
| | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| Bottom | | | | |
| A 305-50T | 0.01546 | 0.02232 | 0.03754 | 0.12724 |
| Old-Style | 0.01809 | 0.03559 | 0.04586 | 0.15217 |
| Top | | | | |
| A 305-50T | 0.02066 | 0.03072 | 0.04855 | 0.31664 |
| Old-Style | 0.02458 | 0.03408 | 0.06737 | 0.12395 |

linear increasing trend from 20,000 to 40,000 psi; however, a 261.23 average percent increase occurs from the 40,000- to the 50,000-psi stress level; whereas, a 152.98 percent increase occurs from the 20,000- to the 40,000-psi stress level.

54. For the first-order interaction effect of reinforcement bar deformation type by stress, the pertinent data are displayed in Table 11 and graphically in Figure 39. Orthogonal comparisons were made of

Table 11
Reinforcement Bar Deformation by Stress Level,
Maximum Crack Width (in.)

| Deformation Type | Stress at | | | |
|------------------|---------------|---------------|---------------|---------------|
| | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| A 305-50T | 0.01806 | 0.02652 | 0.04305 | 0.22194 |
| Old-style | 0.02134 | 0.03484 | 0.05661 | 0.13806 |

BLOCK CHART OF MEANS

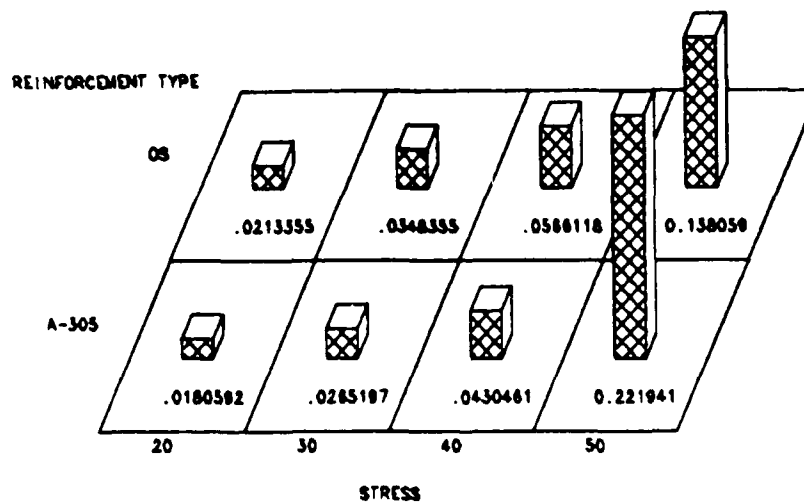


Figure 39. Maximum crack width average over years by reinforcement type

reinforcement bar deformation types within stress levels; the critical difference in maximum crack width was found to be 0.0544 in. As is observed from Table 11, the only difference which exceeds this critical

difference is at the 50,000-psi stress level where the A 305-50T reinforced concrete beams exhibited a significantly larger average maximum crack width than the old-style reinforced concrete beams. One (beam 149) of the four beams which provided data for the top position, A 305-50T deformation type, 50,000-psi stress level treatment combination experienced failure of one of its two reinforcing bars during the winter of 1973-1974 (see Figures 40 and 41 and Appendix A). The loss of approximately one-half of its tensile load-bearing capacity resulted in the



Figure 40. Beam 149 experienced failure of one of two reinforcing bars during winter of 1973-1974

formation of a very large transverse crack which increased in width through subsequent years. Beam 147, one of the four data sources for the top position, old-style deformation type, 50,000-psi stress level treatment combination, experienced the abrupt failure of both reinforcing bars in 1968. This failure completely severed beam 147 and damaged its companion beam, No. 148. Consequently, only two beams (151 and 152) remained for data collection in this treatment condition (top, old-style, 50,000-psi stress). Because of the relatively early failure and



Figure 41. Close-up of beam 149. Note severed rebar

discontinuance of data collection on beams 147 and 148, these data were excluded from the analysis; whereas beam 149, having experienced partial failure, continued to produce crack width data of very large magnitudes. The early failure of the old-style beams and subsequent loss of "incriminating" performance data from the analysis seriously affects the validity of conclusions that might be drawn on the basis of the numbers shown in Tables 10 and 11 concerning deformation type at the 50,000-psi stress level.

55. For the first-order interaction effect of position by stress, the data are displayed in Table 12 and Figure 42. Since these means are based on the same number of observations ($n = 38$), the critical difference between average maximum crack width within stress levels remains at 0.0544 in. Consequently, the only stress level exhibiting a difference larger than 0.0544 in. is the 50,000-psi stress level where the top position exhibits an average maximum crack width of 0.2203 in. and the bottom position exhibits an average maximum crack width of 0.13970 in.

Table 12
Position by Stress,
Maximum Crack Width (in.)

| Position | Stress at | | | |
|----------|---------------|---------------|---------------|---------------|
| | 20,000 psi | 30,000 psi | 40,000 psi | 50,000 psi |
| Bottom | 0.01678 | 0.02895 | 0.04170 | 0.13970 |
| Top | 0.02262 | 0.03240 | 0.05796 | 0.2203 |

BLOCK CHART OF MEANS

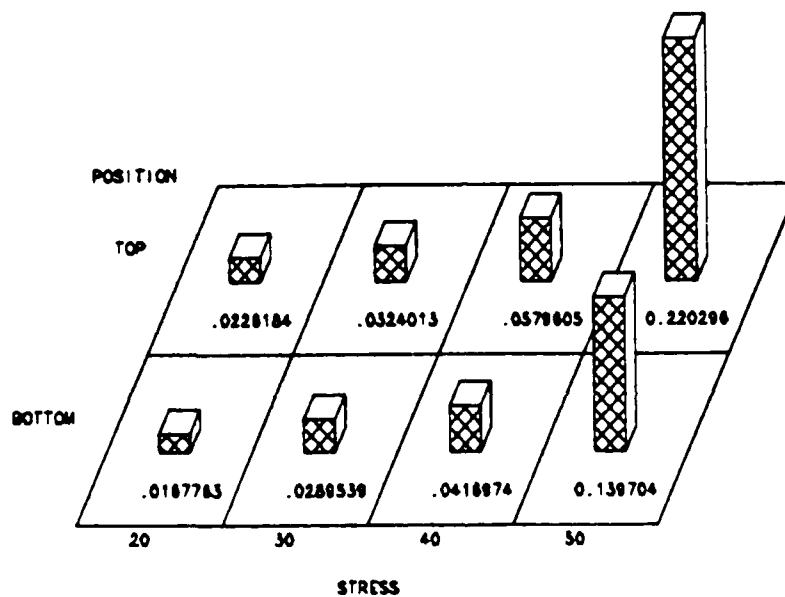


Figure 42. Maximum crack width average
over years by position

56. For the first-order interaction effect of reinforcement bar deformation type and position, the pertinent data are displayed in Table 13. Orthogonal comparisons showed the critical difference between A 305-50T and old-style reinforcement bar deformation to be 0.03866 in. As is observed from Table 13, the two average maximum crack widths which will exceed this critical difference occur at the top position where the A 305-50T exhibited an average maximum crack width of 0.10414 in. and the old-style exhibited a maximum crack width of 0.06249 in. As mentioned

Table 13
Reinforcement Bar Deformation by Position,
Maximum Crack Width (in.)

| | <u>A 305-50T</u> | <u>Old-Style</u> |
|--------|------------------|------------------|
| Bottom | 0.05064 | 0.06293 |
| Top | 0.10414 | 0.06249 |

in the previous discussion of Tables 10 and 11, it is felt that these numbers do not represent actual performance of A 305-50T deformation versus old-style deformation considering the omission of the early failure of old-style beams and subsequent loss of performance data.

57. For the main effect of stress, the pertinent data are exhibited in Table 14. Tukey's w-procedure was used to compare all four means simultaneously. It was found that the average maximum crack widths for stress levels 20,000, 30,000, and 40,000 psi were not significantly different; however, the 50,000-psi stress level exhibited an average maximum crack width which was significantly larger than the 20,000-, 30,000-, and 40,000-psi stress levels.

Table 14
Stress Levels,
Maximum Crack Width (in.)

| <u>20,000</u> | <u>30,000</u> | <u>40,000</u> | <u>50,000</u> |
|---------------|---------------|---------------|---------------|
| <u>psi</u> | <u>psi</u> | <u>psi</u> | <u>psi</u> |
| 0.01970 | 0.03068 | 0.04983 | 0.1800 |

58. For the response variable maximum crack width the data from this investigation indicated that maximum crack width increased linearly for stress levels 20,000, 30,000, and 40,000 psi, and nonlinearly for stress level 50,000 psi. Furthermore, this linear trend when averaged over time showed a marked increase from the 40,000- to the 50,000-psi stress level for all positions and deformation bar types. The largest

increase between the 40,000- and 50,000-psi stress levels occurred for the A 305-50T deformation type with the top position. However, it is felt that the old-style deformation type would have shown a similar trend if data from the beams which failed had been included in the analysis. Beams with the A 305-50T deformation bar type displayed smaller maximum crack widths than the old-style deformation bar type for stress levels of 20,000, 30,000, and 40,000 psi; however, the opposite was true for the 50,000-psi stress level. Again, this reversal in trend is probably due to omission of performance data on the old-style beams which failed.

Linear Models

59. Linear regression analyses of condition, $\%V^2$, and maximum crack width were done for each combination of position, stress, and reinforcement bar deformation over time. The results are shown in Appendix C. The correlation coefficient and the mathematical equation describing each regression line are given for each combination of position, stress, and reinforcement bar deformation over time. In the equations the predictor is the year and the criterion measures are condition rating, $\%V^2$, and maximum crack width.

PART IV: CONCLUSIONS

60. An evaluation of the results of the statistical analysis leads to the following conclusions:

- a. Beams with steel in the bottom-as-cast position deteriorate at a slower rate than do beams with steel in the top-as-cast position for both A 305-50T and old-style deformation type, and beams with steel in the bottom-as-cast position exhibited smaller average maximum crack widths (significant at 50,000-psi stress level).
- b. A 305-50T type reinforcement bar deformation exhibited less severe degradation trends than old-style, and A 305-50T deformation type exhibited a significantly larger $\%V^2$ than did old-style deformation at the 50,000-psi stress level.
- c. As stress levels increased, the conditions of the beams generally decreased and the degradation of $\%V^2$ increased. There were marked increases in maximum crack widths from the 40,000- to 50,000-psi stress levels for all positions and bar deformation types.
- d. At the 50,000-psi stress level the A 305-50T reinforced concrete beams exhibited a significantly larger average maximum crack width than the beams containing old-style deformation bars.*

61. For further clarification of some apparent anomalies, it should be noted here that the zero stress level (control) beams were more difficult to support than the yoked pairs (stressed) of beams. Consequently, they were tossed and moved around during winter storms and became partially covered with sand. The partial covering with sand during most of their exposure time affected a more saturated condition of these beams which resulted in inordinate deterioration due to freezing and thawing. It is felt that this more severe exposure condition adversely affected the performance of the zero stress level beams as reflected in the analysis results of the variable "condition."

* As previously discussed, it is felt that the early failure of one pair of old-style, 50,000-psi stress level beams, and the subsequent loss of "incriminating" performance data seriously affect the validity of conclusions drawn concerning performance of deformation type at the 50,000-psi stress level. With this in mind, and in view of conclusion b above, the data seem to indicate that beams containing bars with A 305-50T deformation performed better than beams containing old-style type bars.

APPENDIX A: EXPOSURE RECORDS OF SPECIMENS

(Revised Aug 1965)

Table 1-TC-B

Section 2

Record of Observation and Testing of Large-beam Tensile Crack Specimens,

Series B, 1954- (Installed Nov 1954)

Beach Row 1

| Beam No. | Nominal Stress psi | Steel Position* | Type** Steel Deformation | 1954-1958 Readings | | | | | | | | | |
|----------|--------------------|-----------------|--------------------------|--------------------|------------------------------|-----------------|-----------|-----------------|-----------|-----------------------------|--------------|------------------|-----------------------------|
| | | | | 0 Cycles, 1954 | | 143 Cycles 1955 | | 310 Cycles 1956 | | 454 Cycles, 1957 | | 525 Cycles, 1958 | |
| | | | | Condition | Pulse Veloc fps ΔV^2 | Condition | Condition | Condition | Condition | Max Crack Width† 1/1000 in. | ΔV^2 | Condition | Max Crack Width† 1/1000 in. |
| 83 | 20,000 | B | A-305 | Sound | 10,890 100 | 100 | 91 | 87 | 173 | 10 | 88 | 173 | 10 |
| 84 | 20,000 | B | A-305 | Sound | 11,150 100 | 100 | 91 | 88 | 168 | 5 | 84 | 170 | 10 |
| 85 | 20,000 | B | OS | Sound | 11,700 100 | 100 | 90 | 84 | 157 | 10 | 84 | 153 | 10 |
| 86 | 20,000 | B | OS | Sound | 11,470 100 | 100 | 87 | 82 | 170 | 10 | 84 | 155 | 10 |
| 87 | 20,000 | B | A-305 | Sound | 10,640 100 | 100 | 90 | 75 | 171 | 5 | 77 | 183 | 5 |
| 88 | 20,000 | B | A-305 | Sound | 10,470 100 | 100 | 84 | 76 | 175 | 10 | 77 | 200 | 10 |
| 89 | 20,000 | B | OS | Sound | 11,255 100 | 100 | 84 | 77 | 162 | 10 | 77 | 167 | 10 |
| 90 | 20,000 | B | OS | Sound | 11,300 100 | 100 | 83 | 83 | 150 | 10 | 77 | 167 | 10 |
| 91 | 30,000 | B | A-305 | Sound | 11,540 100 | 97 | 90 | 79 | 146 | 10 | 75 | 151 | 15 |
| 92 | 30,000 | B | A-305 | Sound | 11,540 100 | 100 | 90 | 82 | 161 | 10 | 81 | 166 | 10 |
| 93 | 30,000 | B | OS | Sound | 12,120 100 | 100 | 87 | 80 | 151 | 15 | 80 | 152 | 15 |
| 94 | 30,000 | B | OS | Sound | 11,605 100 | 100 | 87 | 82 | 145 | 20 | 80 | 169 | 20 |
| 95 | 30,000 | B | A-305 | Sound | 11,905 100 | 100 | 92 | 84 | 154 | 10 | 84 | 156 | 10 |
| 96 | 30,000 | B | A-305 | Sound | 11,195 100 | 100 | 90 | 86 | 162 | 10 | 80 | 174 | 10 |
| 97 | 30,000 | B | OS | Sound | 11,385 100 | 100 | 86 | 86 | 152 | 15 | 86 | 154 | 15 |
| 98 | 30,000 | B | OS | Sound | 11,385 100 | 100 | 90 | 85 | 149 | 20 | 87 | 159 | 20 |
| 99 | 40,000 | B | A-305 | Sound | 10,290 100 | 100 | 88 | 87 | 190 | 15 | 84 | 202 | 15 |
| 100 | 40,000 | B | A-305 | Sound | 10,435 100 | 100 | 88 | 87 | 190 | 10 | 87 | 188 | 15 |
| 101 | 40,000 | B | OS | Sound | 10,400 100 | 98 | 82 | 82 | 195 | 15 | 79 | 191 | 20 |
| 102 | 40,000 | B | OS | Sound | 10,455 100 | 100 | 84 | 82 | 167 | 20 | 81 | 202 | 20 |
| 103 | 40,000 | B | A-305 | Sound | 8,915 100 | 95 | 83 | 80 | 228 | 10 | 77 | 246 | 20 |
| 104 | 40,000 | B | A-305 | Sound | 8,585 100 | 94 | 82 | 72 | 248 | 25 | 74 | 259 | 25 |
| 105 | 40,000 | B | OS | Sound | 9,230 100 | 100 | 86 | 91 | 246 | 10 | 84 | 237 | 20 |
| 106 | 40,000 | B | OS | Sound | 9,435 100 | 100 | 80 | 80 | 236 | 25 | 84 | 238 | 30 |
| 107 | 50,000 | B | A-305 | Sound | 10,310 100 | 100 | 86 | 80 | 195 | 15 | 80 | 191 | 15 |
| 108 | 50,000 | B | A-305 | Sound | 11,385 100 | 98 | 84 | 77 | 147 | 20 | 84 | 154 | 20 |
| 109 | 50,000 | B | OS | Sound | 8,915 100 | 91 | 74 | 72 | 274 | 25 | 80 | 273 | 25 |
| 110 | 50,000 | B | OS | Sound | 10,170 100 | 92 | 74 | 72 | 199 | 25 | 84 | 201 | 25 |
| 111 | 50,000 | B | A-305 | Sound | 9,130 100 | 99 | 79 | 74 | 291 | 20 | 76 | 282 | 25 |
| 112 | 50,000 | B | A-305 | Sound | 9,160 100 | 100 | 86 | 86 | 257 | 25 | 84 | 255 | 25 |
| 113 | 50,000 | B | OS | Sound | 8,850 100 | 93 | 70 | 64 | 243 | 25 | 80 | 270 | 25 |
| 114 | 50,000 | B | OS | Sound | 8,525 100 | 100 | 77 | 77 | 250 | 30 | 84 | 260 | 30 |
| 115 | None | B | A-305 | Sound | 12,985 100 | 96 | 86 | 84 | 115 | 0 | 84 | 122 | 0 |
| 116 | None | B | A-305 | Sound | 13,015 100 | 100 | 88 | 84 | 110 | 0 | 80 | 111 | 0 |
| 117 | None | B | A-305 | Sound | 13,245 100 | 100 | 94 | 94 | 114 | 10 | 84 | 114 | 10 |
| 118 | None | B | OS | Sound | 13,250 100 | 98 | 76 | 69 | 111 | 0 | 84 | 114 | 0 |
| 119 | None | B | OS | Sound | 13,130 100 | 100 | 91 | 90 | 119 | 0 | 84 | 114 | 0 |
| 120 | None | B | OS | Sound | 13,185 100 | 100 | 88 | 89 | 115 | 0 | 84 | 114 | 0 |
| 121 | 20,000 | T | A-305 | Sound | 9,600 100 | 96 | 87 | 80 | 213 | 35 | 84 | 214 | 25 |
| 122 | 20,000 | T | A-305 | Sound | 9,570 100 | 96 | 87 | 80 | 237 | 19 | 84 | 237 | 25 |
| 123 | 20,000 | T | OS | Sound | 9,870 100 | 100 | 86 | 84 | 205 | 10 | 84 | 209 | 10 |
| 124 | 20,000 | T | OS | Sound | 9,675 100 | 100 | 84 | 86 | 216 | 10 | 84 | 214 | 10 |
| 125 | 20,000 | T | A-305 | Sound | 12,960 100 | 100 | 86 | 86 | 120 | 15 | 84 | 121 | 15 |
| 126 | 20,000 | T | A-305 | Sound | 13,160 100 | 100 | 79 | 80 | 122 | 35 | 84 | 127 | 25 |
| 127 | 20,000 | T | OS | Sound | 13,305 100 | 100 | 84 | 80 | 132 | 10 | 84 | 137 | 15 |
| 128 | 20,000 | T | OS | Sound | 13,015 100 | 100 | 92 | 90 | 135 | 10 | 84 | 137 | 15 |
| 129 | 30,000 | T | A-305 | Sound | 9,755 100 | 97 | 84 | 75 | 134 | 15 | 84 | 137 | 15 |
| 130 | 30,000 | T | A-305 | Sound | 9,820 100 | 96 | 84 | 84 | 130 | 20 | 84 | 137 | 20 |
| 131 | 30,000 | T | OS | Sound | 11,675 100 | 96 | 81 | 76 | 136 | 20 | 84 | 137 | 20 |
| 132 | 30,000 | T | OS | Sound | 11,675 100 | 97 | 77 | 76 | 159 | 30 | 80 | 137 | 25 |
| 133 | 30,000 | T | A-305 | Sound | 13,070 100 | 100 | 88 | 84 | 115 | 10 | 84 | 137 | 10 |
| 134 | 30,000 | T | A-305 | Sound | 12,820 100 | 100 | 88 | 84 | 120 | 15 | 84 | 137 | 15 |
| 135 | 30,000 | T | OS | Sound | 12,875 100 | 99 | 87 | 84 | 141 | 15 | 84 | 137 | 15 |
| 136 | 30,000 | T | OS | Sound | 11,340 100 | 96 | 84 | 84 | 136 | 10 | 84 | 137 | 10 |
| 137 | 40,000 | T | A-305 | Sound | 10,510 100 | 94 | 88 | 84 | 136 | 20 | 84 | 137 | 20 |

* 1. Location of beam.

* 2. Location of beam.

** A-305 = Aluminum 305; OS = Other Steel.

† Max crack width is the maximum width of crack measured at the time of testing.

† Max crack width is the maximum width of crack measured at the time of testing.

Table 1-TC-B (Continued)

Beach Row 1

| 1934-1937 Readings | | | | | | | | | | | | | | |
|--------------------|--------------------|----------------|------------------------|----------------|--------------------|--------------|------------------|------------------|------------------|--------------|----------------------------|------------------|--------------|----------------------------|
| Beam No. | Nominal Stress psi | Steel Position | Type Steel Deformation | 0 Cycles, 1954 | | | 143 Cycles, 1955 | 310 Cycles, 1956 | 454 Cycles, 1957 | | | 525 Cycles, 1958 | | |
| | | | | Condition | Pulse Velocity fps | ΔV^2 | Condition | Condition | Condition | ΔV^2 | Max Crack Width 1/1000 in. | Condition | ΔV^2 | Max Crack Width 1/1000 in. |
| 138 | 40,000 | T | A-305 | 100 | 10,490 | 100 | 92 | 94 | 74 | 177 | 25 | 72 | 132 | 25 |
| 139 | 40,000 | T | OS | 100 | 12,095 | 100 | 88 | 74 | 72 | 132 | 14 | 75 | 140 | 15 |
| 140 | 40,000 | T | OS | 100 | 12,225 | 100 | 90 | 74 | 72 | 129 | 19 | 74 | 143 | 15 |
| 141 | 40,000 | T | A-305 | 100 | 9,275 | 100 | 99 | 94 | 74 | 244 | 15 | 70 | 241 | 30 |
| 142 | 40,000 | T | A-305 | 100 | 9,570 | 100 | 100 | 95 | 76 | 223 | 15 | 75 | 237 | 30 |
| 143 | 40,000 | T | OS | 100 | 9,375 | 100 | 94 | 81 | 74 | 224 | 25 | 84 | 234 | 25 |
| 144 | 40,000 | T | OS | 100 | 9,390 | 100 | 95 | 76 | 74 | 231 | 40 | 84 | 238 | 40 |
| 145 | 50,000 | T | A-305 | 100 | 9,435 | 100 | 96 | 82 | 82 | 243 | 40 | 84 | 253 | 40 |
| 146 | 50,000 | T | A-305 | 100 | 9,345 | 100 | 94 | 81 | 72 | 238 | 30 | 81 | 255 | 30 |
| 147 | 50,000 | T | OS | 100 | 8,970 | 100 | 81 | 66 | 62 | 272 | 85 | 72 | 249 | 85 |
| 148 | 50,000 | T | OS | 100 | 8,900 | 100 | 82 | 67 | 67 | 260 | 75 | 70 | 260 | 75 |
| 149 | 50,000 | T | A-305 | 100 | 9,125 | 100 | 99 | 82 | 72 | 225 | 40 | 88 | 235 | 40 |
| 150 | 50,000 | T | A-305 | 100 | 9,175 | 100 | 100 | 82 | 82 | 235 | 25 | 86 | 259 | 30 |
| 151 | 50,000 | T | OS | 100 | 11,130 | 100 | 92 | 80 | 72 | 190 | 15 | 72 | 164 | 15 |
| 152 | 50,000 | T | OS | 100 | 10,655 | 100 | 88 | 72 | 72 | 195 | 25 | 74 | 181 | 25 |
| 153 | None | T | A-305 | 100 | 12,475 | 100 | 94 | 86 | 74 | 120 | 0 | 72 | 121 | 0 |
| 154 | None | T | A-305 | 100 | 12,795 | 100 | 100 | 92 | 87 | 117 | 0 | 88 | 132 | 0 |
| 155 | None | T | A-305 | 100 | 12,375 | 100 | 100 | 90 | 86 | 115 | 0 | 80 | 120 | 0 |
| 156 | None | T | OS | 100 | 13,045 | 100 | 100 | 91 | 90 | 120 | 0 | 82 | 113 | 0 |
| 157 | None | T | OS | 100 | 12,630 | 100 | 98 | 86 | 80 | 120 | 0 | 75 | 124 | 0 |
| 158 | None | T | OS | 100 | 12,710 | 100 | 99 | 78 | 61 | 120 | 10 | 70 | 119 | 15 |

| 1959-1961 Readings | | | | | | | | |
|--------------------|--------------|----------------------------------|------------------|--------------|----------------------------------|------------------|--------------|----------------------------------|
| 575 Cycles, 1959 | | | 745 Cycles, 1960 | | | 887 Cycles, 1961 | | |
| Condi- tion | ΔV^2 | Max Crack Width 1/1000 in. | Condi- tion†† | ΔV^2 | Max Crack Width 1/1000 in. | Condi- tion | ΔV^2 | Max Crack Width 1/1000 in. |
| 84 | 161 | 15 | 84 | 180 | 10 | 75 | 177 | 10 |
| 96 | 161 | 20 | 88 | 199 | 10 | 77 | 179 | 10 |
| 91 | 143 | 15 | 91 | 193 | 10 | 79 | 180 | 10 |
| 82 | 150 | 15 | 82 | 137 | 10 | 67 | 172 | 10 |
| 72 | 176 | 15 | 72 | 143 | 10 | 61 | 181 | 10 |
| 73 | 181 | 10 | 73 | 133 | 10 | 62 | 174 | 10 |
| 86 | 160 | 15 | 96 | 182 | 10 | 73 | 177 | 10 |
| 73 | 153 | 10 | 78 | 147 | 10 | 74 | 171 | 10 |
| 80 | 140 | 15 | 80 | 197 | 10 | 71 | 170 | 10 |
| 89 | 154 | 10 | 82 | 162 | 10 | 67 | 172 | 10 |
| 80 | 152 | 30 | 80 | 163 | 20 | 71 | 171 | 10 |
| 75 | 153 | 25 | 75 | 113 | 30 | 69 | 177 | 10 |
| 85 | 144 | 25 | 85 | 163 | 15 | 70 | 172 | 10 |
| 85 | 162 | 20 | 85 | 167 | 15 | 69 | 173 | 10 |
| 73 | 147 | 20 | 73 | 132 | 10 | 69 | 168 | 10 |
| 79 | 145 | 20 | 79 | 161 | 15 | 64 | 174 | 10 |
| 77 | 139 | 25 | 77 | 174 | 15 | 67 | 175 | 10 |
| 72 | 179 | 25 | 72 | 197 | 15 | 72 | 176 | 10 |
| 70 | 191 | 35 | 70 | 131 | 20 | 65 | 177 | 10 |
| 69 | 133 | 45 | 69 | 137 | 20 | 67 | 130 | 10 |
| 72 | 233 | 25 | 72 | 137 | 15 | 61 | 184 | 10 |
| 66 | 230 | 25 | 66 | 217 | 10 | 67 | 183 | 10 |
| 66 | 234 | 20 | 66 | 161 | 15 | 67 | 179 | 10 |
| 66 | 233 | 21 | 66 | 183 | 15 | 67 | 180 | 10 |
| 66 | 195 | 20 | 66 | 177 | 15 | 67 | 180 | 10 |
| 72 | 140 | 25 | 72 | 177 | 20 | 67 | 176 | 10 |
| 72 | 154 | 25 | 72 | 177 | 20 | 67 | 171 | 10 |
| 72 | 147 | 20 | 72 | 174 | 20 | 67 | 174 | 10 |
| 72 | 144 | 20 | 72 | 167 | 20 | 67 | 173 | 10 |
| 72 | 147 | 20 | 72 | 177 | 20 | 67 | 173 | 10 |

$$\left(\frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} \right)$$

(Revised Aug 1965)

Table 1-TC-B (Continued)

Section 2

| Beach Row 1 | | | | | | | | | | | | |
|-------------|--------------------|----------------|------------------------|--------------------|-----------------|----------------------------|------------------|-----------------|----------------------------|------------------|-----------------|----------------------------|
| Beam No. | Nominal Stress psi | Steel Position | Type Steel Deformation | 1955-1961 Readings | | | | | | 507 Cycles, 1961 | | |
| | | | | 507 Cycles, 1959 | | | 507 Cycles, 1960 | | | Condition | AV ² | Max Crack Width 1/1000 in. |
| | | | | Condition | AV ² | Max Crack Width 1/1000 in. | Condition | AV ² | Max Crack Width 1/1000 in. | | | |
| 113 | 50,000 | B | 08 | 65 | 238 | 30 | 65 | 173 | 25 | 65 | 229 | 30 |
| 114 | 50,000 | B | 08 | 65 | 235 | 35 | 65 | 172 | 25 | 65 | 225 | 30 |
| 115 | None | B | A-305 | 78 | 113 | 0 | 78 | 112 | 0 | 69 | 121 | 0 |
| 116 | None | B | A-305 | 80 | 107 | 0 | 80 | 114 | 0 | 70 | 117 | 0 |
| 117 | None | B | A-305 | 80 | 107 | 30 | 80 | 109 | 10 | 67 | 117 | 10 |
| 118 | None | B | 08 | 55 | 111 | 0 | 55 | 112 | 0 | 60 | 124 | 0 |
| 119 | None | B | 08 | 70 | 104 | 0 | 70 | 111 | 0 | 67 | 123 | 0 |
| 120 | None | B | 08 | 59 | 105 | 0 | 59 | 112 | 0 | 66 | 119 | 0 |
| 121 | 20,000 | T | A-305 | 80 | 201 | 15 | 80 | 165 | 10 | 77 | 230 | 10 |
| 122 | 20,000 | T | A-305 | 80 | 212 | 15 | 80 | 166 | 10 | 70 | 229 | 10 |
| 123 | 20,000 | T | 08 | 77 | 216 | 20 | 77 | 173 | 10 | 65 | 241 | 10 |
| 124 | 20,000 | T | 08 | 57 | 17 | 30 | 57 | 176 | 15 | 65 | 219 | 10 |
| 125 | 20,000 | T | A-305 | 78 | 174 | 20 | 78 | 173 | 10 | 65 | 241 | 10 |
| 126 | 20,000 | T | A-305 | 77 | 113 | 10 | 77 | 173 | 10 | 65 | 241 | 10 |
| 127 | 20,000 | T | 08 | 66 | 121 | 15 | 66 | 173 | 10 | 65 | 241 | 10 |
| 128 | 20,000 | T | 08 | 82 | 130 | 20 | 82 | 133 | 10 | 75 | 241 | 10 |
| 129 | 30,000 | T | A-305 | 65 | 214 | 15 | 65 | 173 | 10 | 65 | 241 | 10 |
| 130 | 30,000 | T | A-305 | 81 | 213 | 15 | 81 | 173 | 10 | 65 | 241 | 10 |
| 131 | 30,000 | T | 08 | 75 | 144 | 15 | 75 | 174 | 15 | 65 | 241 | 10 |
| 132 | 30,000 | T | 08 | 72 | 147 | 15 | 72 | 175 | 15 | 62 | 241 | 10 |
| 133 | 30,000 | T | A-305 | 77 | 119 | 20 | 77 | 173 | 15 | 65 | 241 | 10 |
| 134 | 30,000 | T | A-305 | 77 | 122 | 25 | 77 | 173 | 15 | 65 | 241 | 10 |
| 135 | 30,000 | T | 08 | 73 | 134 | 30 | 73 | 173 | 20 | 65 | 241 | 10 |
| 136 | 30,000 | T | 08 | 77 | 143 | 25 | 77 | 173 | 20 | 65 | 241 | 10 |
| 137 | 40,000 | T | A-305 | 54 | 173 | 25 | 54 | 164 | 20 | 65 | 241 | 10 |
| 138 | 40,000 | T | A-305 | 68 | 171 | 40 | 68 | 133 | 30 | 65 | 241 | 10 |
| 139 | 40,000 | T | 08 | 75 | 139 | 35 | 75 | 145 | 30 | 65 | 241 | 10 |
| 140 | 40,000 | T | 08 | 70 | 139 | 30 | 70 | 149 | 15 | 65 | 241 | 10 |
| 141 | 40,000 | T | A-305 | 68 | 231 | 25 | 68 | 213 | 25 | 65 | 241 | 10 |
| 142 | 40,000 | T | A-305 | 73 | 208 | 20 | 73 | 173 | 15 | 65 | 241 | 10 |
| 143 | 40,000 | T | 08 | 66 | 223 | 20 | 66 | 169 | 15 | 65 | 241 | 10 |
| 144 | 40,000 | T | 08 | 75 | 226 | 25 | 75 | 171 | 20 | 65 | 241 | 10 |
| 145 | 50,000 | T | A-305 | 70 | 235 | 25 | 70 | 170 | 20 | 65 | 241 | 10 |
| 146 | 50,000 | T | A-305 | 65 | 235 | 30 | 65 | 171 | 20 | 65 | 241 | 10 |
| 147 | 50,000 | T | 08 | 61 | 132 | 105 | 61 | 221 | 100 | 65 | 241 | 10 |
| 148 | 50,000 | T | 08 | 64 | 242 | 110 | 64 | 258 | 90 | 65 | 241 | 10 |
| 149 | 50,000 | T | A-305 | 70 | 243 | 30 | 70 | 164 | 15 | 65 | 241 | 10 |
| 150 | 50,000 | T | A-305 | 66 | 240 | 35 | 66 | 173 | 25 | 65 | 241 | 10 |
| 151 | 50,000 | T | 08 | 64 | 159 | 30 | 64 | 176 | 25 | 65 | 241 | 10 |
| 152 | 50,000 | T | 08 | 61 | 173 | 35 | 61 | 197 | 30 | 65 | 241 | 10 |
| 153 | None | T | A-305 | 55 | 116 | 0 | 55 | 123 | 0 | 65 | 241 | 10 |
| 154 | None | T | A-305 | 59 | 114 | 0 | 59 | 123 | 0 | 65 | 241 | 10 |
| 155 | None | T | A-305 | 77 | 116 | 0 | 77 | 123 | 0 | 65 | 241 | 10 |
| 156 | None | T | 08 | 78 | 111 | 0 | 78 | 117 | 0 | 65 | 241 | 10 |
| 157 | None | T | 08 | 77 | 115 | 0 | 77 | 123 | 0 | 65 | 241 | 10 |
| 158 | None | T | 08 | 66 | 117 | 20 | 66 | 126 | 15 | 65 | 241 | 10 |

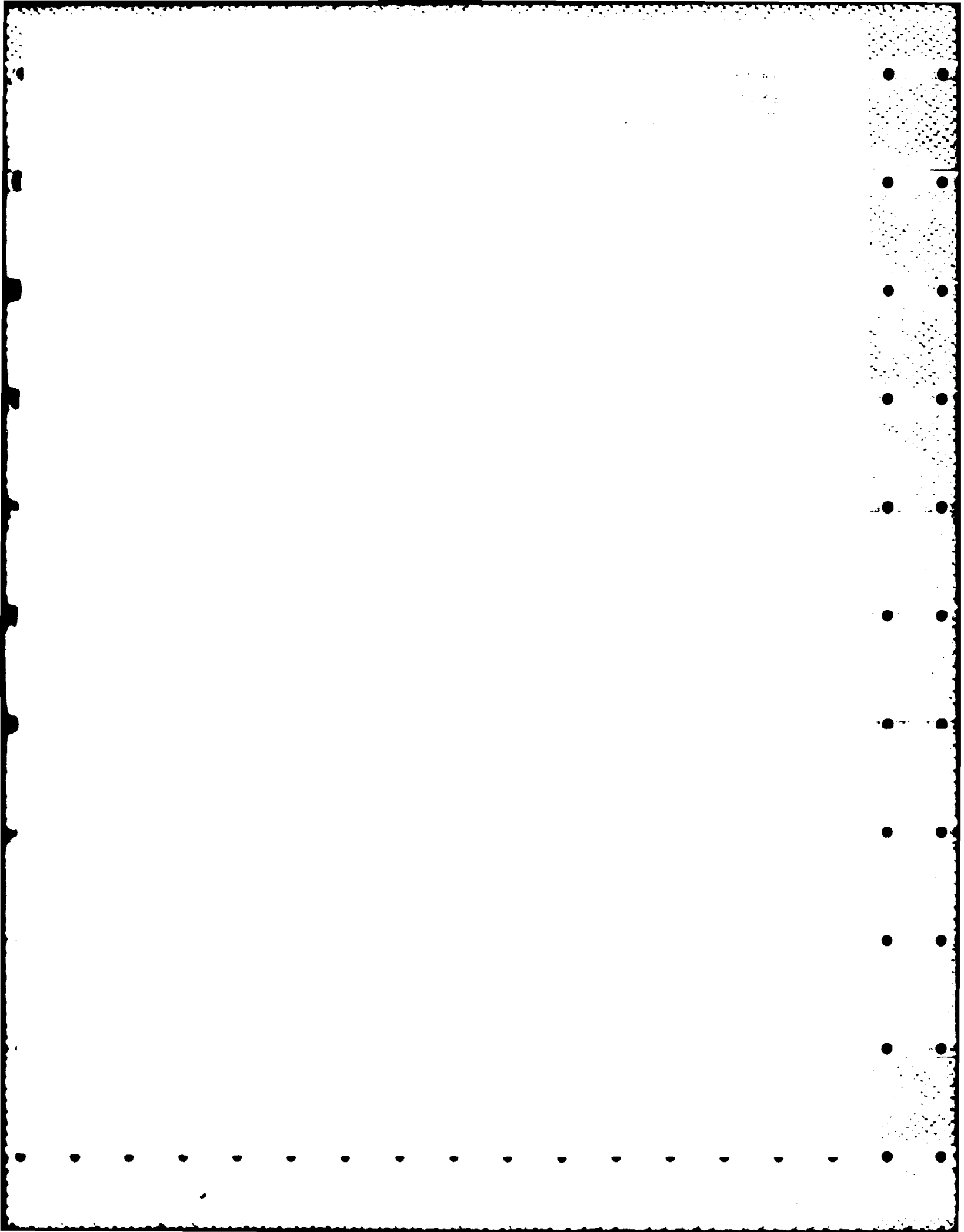
Beach Row 1

| 1955-1961 Readings | | | | | | | | | | | |
|--------------------|-----------------|----------------------------|------------------|-----------------|----------------------------|------------------|-----------------|----------------------------|------------------|-----------------|----------------------------|
| 507 Cycles, 1959 | | | 507 Cycles, 1960 | | | 507 Cycles, 1961 | | | 507 Cycles, 1962 | | |
| Beam No. | AV ² | Max Crack Width 1/1000 in. | Beam No. | AV ² | Max Crack Width 1/1000 in. | Beam No. | AV ² | Max Crack Width 1/1000 in. | Beam No. | AV ² | Max Crack Width 1/1000 in. |
| | | | | | | | | | | | |
| 84 | 141 | 10 | 64 | 131 | 20 | 64 | 117 | 10 | 64 | 117 | 10 |
| 85 | 126 | 10 | 71 | 145 | 20 | 64 | 117 | 10 | 64 | 117 | 10 |
| 86 | 142 | 15 | 77 | 141 | 30 | 64 | 117 | 30 | 64 | 117 | 30 |
| 87 | 139 | 10 | 74 | 112 | 25 | 64 | 117 | 25 | 64 | 117 | 25 |
| 88 | 175 | 10 | 68 | 127 | 20 | 64 | 117 | 20 | 64 | 117 | 20 |
| 89 | 134 | 10 | 61 | 147 | 15 | 64 | 117 | 15 | 64 | 117 | 15 |
| 90 | 135 | 15 | 71 | 114 | 20 | 64 | 117 | 20 | 64 | 117 | 20 |
| 91 | 171 | 15 | 61 | 130 | 20 | 64 | 117 | 20 | 64 | 117 | 20 |
| 92 | 164 | 15 | 77 | 119 | 30 | 64 | 117 | 30 | 64 | 117 | 30 |
| 93 | 139 | 15 | 74 | 138 | 25 | 64 | 117 | 25 | 64 | 117 | 25 |

(Revised Aug 1965)

Table 1-TC-B (Continued)

| Beam No. | Nominal Stress psi | Steel Position | Type Steel Deformation | 970 Cycles, 1962 | | 1080 Cycles, 1962 | | 1080 Cycles, 1962 | | 1080 Cycles, 1962 | |
|----------|--------------------|----------------|------------------------|------------------|--------------------------------------|-------------------|--------------------------------------|-------------------|--------------------------------------|-------------------|--------------------------------------|
| | | | | Condition | Max Crack Width $\frac{1}{1000}$ in. | Condition | Max Crack Width $\frac{1}{1000}$ in. | Condition | Max Crack Width $\frac{1}{1000}$ in. | Condition | Max Crack Width $\frac{1}{1000}$ in. |
| | | | | | | | | | | | |
| 93 | 30,000 | B | OS | 71 | 136 | 20 | 70 | 100 | 20 | 70 | 100 |
| 94 | 30,000 | B | OS | 69 | 163 | 25 | 69 | 133 | 25 | 69 | 133 |
| 95 | 30,000 | B | A-305 | 79 | 146 | 15 | 94 | 107 | 15 | 94 | 107 |
| 96 | 30,000 | B | A-305 | 69 | 162 | 20 | 81 | 120 | 20 | 81 | 120 |
| 97 | 30,000 | B | OS | 64 | 147 | 10 | 64 | 104 | 10 | 64 | 104 |
| 98 | 30,000 | B | OS | 64 | 154 | 15 | 67 | 111 | 15 | 67 | 111 |
| 99 | 40,000 | B | A-305 | 62 | 140 | 15 | 67 | 130 | 15 | 67 | 130 |
| 100 | 40,000 | B | A-305 | 64 | 212 | 15 | 75 | 143 | 15 | 75 | 143 |
| 101 | 40,000 | B | OS | 64 | 184 | 25 | 68 | 148 | 25 | 68 | 148 |
| 102 | 40,000 | B | OS | 67 | 206 | 20 | 66 | 143 | 20 | 66 | 143 |
| 103 | 40,000 | B | A-305 | 59 | 210 | 15 | 73 | 147 | 15 | 73 | 147 |
| 104 | 40,000 | B | A-305 | 66 | 258 | 25 | 67 | 185 | 25 | 67 | 185 |
| 105 | 40,000 | B | OS | 72 | 358 | 20 | 73 | 147 | 20 | 73 | 147 |
| 106 | 40,000 | B | OS | 70 | 344 | 15 | 66 | 140 | 15 | 66 | 140 |
| 107 | 50,000 | B | A-305 | 54 | 194 | 15 | 67 | 141 | 15 | 67 | 141 |
| 108 | 50,000 | B | A-305 | 59 | 134 | 15 | 67 | 147 | 15 | 67 | 147 |
| 109 | 50,000 | B | OS | 61 | 192 | 25 | 66 | 140 | 25 | 66 | 140 |
| 110 | 50,000 | B | OS | 59 | 197 | 25 | 67 | 147 | 25 | 67 | 147 |
| 111 | 50,000 | B | A-305 | 57 | 206 | 25 | 67 | 147 | 25 | 67 | 147 |
| 112 | 50,000 | B | A-305 | 66 | 267 | 15 | 67 | 147 | 15 | 67 | 147 |
| 113 | 50,000 | B | OS | 67 | 204 | 15 | 67 | 147 | 15 | 67 | 147 |
| 114 | 50,000 | B | OS | 67 | 206 | 15 | 67 | 147 | 15 | 67 | 147 |
| 115 | None | B | A-305 | 66 | 113 | 0 | 67 | 147 | 0 | 67 | 147 |
| 116 | None | B | A-305 | 70 | 112 | 0 | 67 | 147 | 0 | 67 | 147 |
| 117 | None | B | A-305 | 67 | 120 | 10 | 67 | 147 | 10 | 67 | 147 |
| 118 | None | B | OS | 67 | 96 | 0 | 67 | 147 | 0 | 67 | 147 |
| 119 | None | B | OS | 69 | 104 | 0 | 67 | 147 | 0 | 67 | 147 |
| 120 | None | H | OS | 67 | 114 | 0 | 67 | 147 | 0 | 67 | 147 |
| 121 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 122 | 30,000 | T | A-305 | 67 | 107 | 10 | 67 | 147 | 10 | 67 | 147 |
| 123 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 124 | 30,000 | T | OS | 67 | 104 | 10 | 67 | 147 | 10 | 67 | 147 |
| 125 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 126 | 30,000 | T | A-305 | 67 | 110 | 10 | 67 | 147 | 10 | 67 | 147 |
| 127 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 128 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 129 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 130 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 131 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 132 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 133 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 134 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 135 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 136 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 137 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 138 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 139 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 140 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 141 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 142 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 143 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 144 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 145 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 146 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 147 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 148 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 149 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 150 | 30,000 | T | A-305 | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 151 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 152 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 153 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 154 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 155 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 156 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 157 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 158 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 159 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 160 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 161 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |
| 162 | 30,000 | T | OS | 67 | 106 | 10 | 67 | 147 | 10 | 67 | 147 |



(Revised Jan 1973)

Table 1-PC-8 (Continued)

Section 2

Beam No. 1

| 1969-1972 Readings | | | | | | | | | | | | |
|--------------------|---------------------|---------------------|--------------------------|-------------------|----------------------------|-------------------|----------------------------|-------------------|--------------------|--------------------|--------------------|--------------------|
| Beam No. | Nominal Stress, psi | Steel Reinforcement | Type Steel Reinforcement | 1360 Cycles, 1969 | | 1510 Cycles, 1970 | | 1969 Cycles, 1971 | | 1969 Cycles, 1972 | | After Re-load, in. |
| | | | | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Before Unload, in. | After Re-load, in. | Before Unload, in. | |
| | | | | | $\sqrt{f'}$ | | $\sqrt{f'}$ | | Not loaded | | Not loaded | After Re-load, in. |
| 133 | 30,000 | T | A-305 | * | 50 | 15 | 63 | ** | 82 | 52 | 16 | 15 |
| 134 | 30,000 | T | A-305 | * | 54 | 15 | 72 | * | 84 | 54 | 16 | 16 |
| 135 | 30,000 | T | OS | * | 74 | 15 | 77 | * | 65 | 44 | 30 | 30 |
| 136 | 30,000 | T | OS | * | 89 | 25 | 65 | * | 79 | 58 | 30 | 35 |
| 137 | 30,000 | T | A-305 | * | 73 | 40 | 51 | * | 51 | 72 | 50 | 70 |
| 138 | 40,000 | T | A-305 | * | 84 | 40 | 57 | * | 98 | 47 | 50 | 75 |
| 139 | 40,000 | T | OS | * | 57 | 30 | 67 | * | 70 | 51 | 30 | 35 |
| 140 | 40,000 | T | OS | * | 70 | 20 | 64 | * | 74 | 50 | 30 | 35 |
| 141 | 40,000 | T | A-305 | * | 106 | 30 | 55 | * | 155 | 63 | 30 | 35 |
| 142 | 40,000 | T | A-305 | * | 116 | 30 | 59 | * | 135 | 73 | 30 | 35 |
| 143 | 40,000 | T | OS | * | 144 | 25 | 65 | * | 119 | 111 | 35 | 35 |
| 144 | 40,000 | T | OS | * | 112 | 35 | 62 | * | 109 | 95 | 45 | 45 |
| 145 | 40,000 | T | A-305 | * | 139 | 30 | 70 | * | 103 | 88 | 45 | 70 |
| 146 | 40,000 | T | A-305 | * | 120 | 30 | 53 | * | 114 | 75 | 40 | 45 |
| 147 | 40,000 | T | OS | * | 98 | 100 | 59 | * | 162 | 48 | 115 | 125 |
| 148 | 50,000 | T | OS | * | 111 | 55 | 64 | * | 157 | 147 | 100 | 120 |
| 149 | 50,000 | T | A-305 | * | 147 | 35 | 64 | * | 124 | 60 | 45 | 50 |
| 150 | 50,000 | T | A-305 | * | 190 | 40 | 65 | * | 145 | 60 | 30 | 30 |
| 151 | 50,000 | T | OS | * | 71 | 35 | 57 | * | 111 | 64 | 45 | 45 |
| 152 | 50,000 | T | OS | * | 82 | 35 | 54 | * | 102 | 70 | 40 | 45 |
| 153 | None | T | A-305 | * | 6 | 0 | 44 | * | 74 | | | |
| 154 | None | T | A-305 | * | 75 | 0 | 77 | * | 73 | | | |
| 155 | None | T | A-305 | * | 67 | 0 | 67 | * | 70 | | | |
| 156 | None | T | OS | * | 74 | 0 | 66 | * | 69 | | | |
| 157 | None | T | OS | * | 64 | 0 | 60 | * | 87 | | | |
| 158 | None | T | OS | * | 53 | 15 | 52 | * | 68 | | 10 | |

Beam No. 1

| 1969-1972 Readings | | | | | | | | | | | | |
|--------------------|---------------------|---------------------|--------------------------|-------------------|----------------------------|-------------------|----------------------------|-------------------|--------------------|--------------------|--------------------|--------------------|
| Beam No. | Nominal Stress, psi | Steel Reinforcement | Type Steel Reinforcement | 1360 Cycles, 1969 | | 1510 Cycles, 1970 | | 1969 Cycles, 1971 | | 1969 Cycles, 1972 | | After Re-load, in. |
| | | | | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Before Unload, in. | After Re-load, in. | Before Unload, in. | |
| | | | | | $\sqrt{f'}$ | | $\sqrt{f'}$ | | Not loaded | | Not loaded | |
| 159 | 30,000 | S | A-305 | * | 45 | 15 | 52 | 15 | 59 | 60 | 15 | 15 |
| 160 | 30,000 | S | A-305 | * | 68 | 15 | 68 | 15 | 67 | 71 | 20 | 20 |
| 161 | 30,000 | S | OS | * | 72 | 20 | 72 | 20 | 71 | 53 | 25 | 25 |
| 162 | 30,000 | S | OS | * | 85 | 15 | 66 | 15 | 64 | 57 | 20 | 20 |
| 163 | 30,000 | S | A-305 | * | 50 | 30 | 53 | 30 | 52 | 69 | 35 | 35 |
| 164 | 40,000 | S | A-305 | * | 50 | 20 | 64 | 20 | 59 | 70 | 35 | 35 |
| 165 | 40,000 | S | OS | * | 69 | 20 | 67 | 20 | 69 | 73 | 30 | 30 |
| 166 | 40,000 | S | OS | * | 73 | 10 | 64 | 10 | 65 | 61 | 25 | 25 |
| 167 | 40,000 | S | A-305 | * | 73 | 35 | 70 | 35 | 70 | 65 | 25 | 25 |
| 168 | 40,000 | S | A-305 | * | 70 | 20 | 67 | 20 | 67 | 55 | 20 | 20 |
| 169 | 40,000 | S | OS | * | 78 | 40 | 66 | 40 | 64 | 61 | 40 | 40 |
| 170 | 40,000 | S | OS | * | 64 | 75 | 66 | 35 | 67 | 61 | 35 | 35 |
| 171 | 40,000 | S | A-305 | * | 77 | 100 | 72 | 45 | 68 | 60 | 40 | 40 |
| 172 | 40,000 | S | A-305 | * | 68 | 20 | 67 | 45 | 66 | 68 | 35 | 35 |
| 173 | 40,000 | S | OS | * | 60 | 70 | 64 | 30 | 61 | 61 | 35 | 35 |
| 174 | 40,000 | S | OS | * | 47 | 35 | 62 | 35 | 61 | 64 | 35 | 35 |
| 175 | 40,000 | S | A-305 | * | 67 | 35 | 60 | 35 | 59 | 58 | 35 | 35 |
| 176 | 40,000 | S | A-305 | * | 64 | 70 | 64 | 35 | 61 | 60 | 35 | 35 |
| 177 | 40,000 | S | OS | * | 67 | 30 | 64 | 30 | 61 | 64 | 35 | 35 |
| 178 | 40,000 | S | OS | * | 65 | 30 | 64 | 35 | 64 | 61 | 35 | 35 |
| 179 | 40,000 | S | A-305 | * | 71 | 70 | 64 | 35 | 62 | 60 | 35 | 35 |
| 180 | 40,000 | S | A-305 | * | 65 | 35 | 61 | 35 | 60 | 60 | 35 | 35 |
| 181 | 40,000 | S | OS | * | 71 | 45 | 61 | 35 | 68 | 67 | 40 | 40 |
| 182 | 40,000 | S | OS | * | 81 | 50 | 64 | 35 | 64 | 60 | 35 | 35 |
| 183 | 40,000 | S | A-305 | * | 60 | 40 | 59 | 35 | 64 | 60 | 35 | 35 |

- * The condition of specimens was not rated by panel of observers.
- † Cracks in concrete below readings were not obtained in 1969 due to malfunction of testing equipment.
- ‡ Cracks in concrete above readings were not obtained in 1969 due to malfunction of testing equipment.
- § Cracks in concrete above and below readings were not obtained in 1969 due to malfunction of testing equipment.
- ¶ Cracks in concrete above and below readings were not obtained in 1969 due to malfunction of testing equipment.
- ‡ Cracks in concrete above and below readings were not obtained in 1969 due to malfunction of testing equipment.

(Revised Jan 1973)

Table 1-TC-B (Concluded)

Section 2

Beam Row 1

| Table 1. Test Results | | | | | | | | | | | | | | | |
|-----------------------|--------------------|----------------|------------------------|-----------------------|----------------------------|---------------------|----------------------------|---------------------|----------------------------|---------------------|----------------------------|---------------------|----------------------------|----|----|
| Table 1. Test Results | | | | | | | | | | | | | | | |
| Beam No. | Nominal Stress psi | Steel Position | Type Steel Deformation | Table 1. Test Results | | | | | | | | | | | |
| | | | | 1861 Cycles, 1st | | 2005 Cycles, 1st | | 2158 Cycles, 1st | | 2347 Cycles, 1st | | 2444 Cycles, 1st | | | |
| | | | | Con- di- tion | Max Crack Width 1/1000 in. | Con- di- tion | Max Crack Width 1/1000 in. | Con- di- tion | Max Crack Width 1/1000 in. | Con- di- tion | Max Crack Width 1/1000 in. | Con- di- tion | Max Crack Width 1/1000 in. | | |
| 108 | 50,000 | B | A-305 | 57 | 73 | 40 | 57 | 26 | 45 | 57 | 41 | 41 | 57 | 41 | 41 |
| 109 | 50,000 | B | OS | 66 | 144 | 50 | 66 | 50 | 50 | 66 | 49 | 55 | 66 | 41 | 55 |
| 110 | 50,000 | B | OS | 55 | 117 | 50 | 55 | 56 | 50 | 55 | 95 | 55 | 55 | 81 | 50 |
| 111 | 50,000 | B | A-305 | 53 | 91 | 50 | 55 | 52 | 50 | 56 | 49 | 50 | 51 | 50 | 50 |
| 112 | 50,000 | B | A-305 | 63 | 93 | 40 | 63 | 50 | 45 | 63 | 70 | 45 | 63 | 70 | 45 |
| 113 | 50,000 | B | OS | 49 | 102 | 55 | 49 | 53 | 55 | 48 | 92 | 60 | 48 | 86 | 60 |
| 114 | 50,000 | B | OS | 53 | 180 | 55 | 52 | 42 | 65 | 52 | 64 | 65 | 50 | 62 | 50 |
| 115 | None | B | A-305 | 59 | 68 | 0 | 54 | 26 | 0 | 58 | 43 | 0 | 56 | 49 | 0 |
| 116 | None | B | A-305 | 65 | 58 | 0 | 67 | 25 | 0 | 67 | 28 | 0 | 66 | 29 | 0 |
| 117 | None | B | A-305 | 65 | 68 | 0 | 65 | 28 | 0 | 65 | 43 | 0 | 67 | 40 | 0 |
| 118 | None | B | OS | 40 | 66 | 0 | 39 | 27 | 0 | 43 | 43 | 0 | 44 | 41 | 0 |
| 119 | None | B | OS | 52 | 67 | 0 | 53 | 30 | 0 | 56 | 49 | 0 | 52 | 47 | 0 |
| 120 | None | B | OS | 54 | 69 | 0 | 56 | 27 | 0 | 56 | 43 | 0 | 55 | 41 | 0 |
| 121 | 20,000 | T | A-305 | 57 | 41 | 25 | 74 | 49 | 25 | 74 | 79 | 25 | 71 | 73 | 25 |
| 122 | 20,000 | T | A-305 | 82 | 55 | 25 | 70 | 52 | 30 | 69 | 81 | 25 | 67 | 77 | 20 |
| 123 | 20,000 | T | OS | 72 | 119 | 30 | 61 | 47 | 35 | 61 | 76 | 40 | 61 | 74 | 35 |
| 124 | 20,000 | T | OS | 71 | 111 | 35 | 65 | 59 | 40 | 65 | 98 | 40 | 65 | 92 | 45 |
| 125 | 20,000 | T | A-305 | 64 | 45 | 20 | 60 | 34 | 20 | 59 | 59 | 25 | 58 | 53 | 25 |
| 126 | 20,000 | T | A-305 | 77 | 63 | 20 | 77 | 34 | 25 | 76 | 60 | 25 | 77 | 54 | 25 |
| 127 | 20,000 | T | OS | 76 | 68 | 10 | 69 | 28 | 10 | 70 | 42 | 10 | 68 | 37 | 20 |
| 128 | 20,000 | T | OS | 69 | 69 | 5 | 69 | 27 | 5 | 70 | 47 | 10 | 68 | 43 | 15 |
| 129 | 30,000 | T | A-305 | 60 | 40 | 20 | 55 | 50 | 25 | 55 | 77 | 30 | 53 | 73 | 35 |
| 130 | 30,000 | T | A-305 | 71 | 84 | 35 | 66 | 45 | 35 | 65 | 69 | 35 | 62 | 66 | 35 |
| 131 | 30,000 | T | OS | 74 | 56 | 50 | 73 | 34 | 50 | 74 | 50 | 50 | 72 | 47 | 50 |
| 132 | 30,000 | T | OS | 61 | 84 | 40 | 61 | 35 | 50 | 61 | 54 | 50 | 58 | 51 | 50 |
| 133 | 30,000 | T | A-305 | 63 | 45 | 20 | 63 | 26 | 25 | 63 | 44 | 30 | 61 | 42 | 30 |
| 134 | 30,000 | T | A-305 | 66 | 61 | 25 | 66 | 28 | 25 | 66 | 47 | 25 | 64 | 45 | 25 |
| 135 | 30,000 | T | OS | 62 | 53 | 35 | 61 | 28 | 30 | 61 | 44 | 30 | 61 | 42 | 35 |
| 136 | 30,000 | T | OS | 65 | 53 | 35 | 63 | 33 | 30 | 64 | 50 | 25 | 61 | 48 | 25 |
| 137 | 40,000 | T | A-305 | 51 | 89 | 70 | 49 | ## | 70 | 49 | 50 | 70 | 49 | 50 | 70 |
| 138 | 40,000 | T | A-305 | 57 | 106 | 70 | 56 | 37 | 70 | 56 | 60 | 75 | 57 | 56 | 70 |
| 139 | 40,000 | T | OS | 67 | 92 | 35 | 67 | 26 | 40 | 66 | 29 | 45 | 66 | 38 | 45 |
| 140 | 40,000 | T | OS | 69 | 82 | 35 | 70 | 29 | 40 | 68 | 53 | 40 | 67 | 49 | 40 |
| 141 | 40,000 | T | A-305 | 55 | 99 | 55 | 54 | 52 | 55 | 54 | 91 | 60 | 54 | 84 | 60 |
| 142 | 40,000 | T | A-305 | 59 | 111 | 50 | 58 | 50 | 55 | 57 | 80 | 60 | 57 | 80 | 60 |
| 143 | 40,000 | T | OS | 65 | 116 | 30 | 64 | 48 | 35 | 64 | 74 | 35 | 64 | 71 | 30 |
| 144 | 40,000 | T | OS | 61 | 115 | 55 | 61 | 51 | 60 | 61 | 70 | 65 | 61 | 73 | 70 |
| 145 | 50,000 | T | A-305 | 68 | 101 | 65 | 56 | 50 | 70 | 65 | 46 | 65 | 56 | 79 | 65 |
| 146 | 50,000 | T | A-305 | 52 | 98 | 50 | 51 | 68 | 55 | 49 | 116 | 55 | 48 | 96 | 45 |
| 147 | 50,000 | T | OS | Failed§§ | | | | | | | | | | | |
| 148 | 50,000 | T | OS | Damaged §§ | | | | | | | | | | | |
| 149 | 50,000 | T | A-305 | 68 | 126 | 45 | 68 | 50 | 55 | 68 | 93 | 60 | 67 | 77 | 55 |
| 150 | 50,000 | T | A-305 | 65 | 142 | 50 | 65 | 62 | 50 | 64 | 46 | 50 | 66 | 80 | 50 |
| 151 | 50,000 | T | OS | 57 | 79 | 50 | 58 | 30 | 50 | 57 | 49 | 55 | 57 | 45 | 55 |
| 152 | 50,000 | T | OS | 54 | 111 | 50 | 54 | 40 | 55 | 54 | 66 | 55 | 54 | 63 | 50 |
| 153 | None | T | A-305 | 44 | 70 | 0 | 44 | 30 | 0 | 45 | 48 | 0 | 43 | 45 | 0 |
| 154 | None | T | A-305 | 61 | 64 | 0 | 62 | 30 | 0 | 62 | 45 | 0 | 55 | 44 | 0 |
| 155 | None | T | A-305 | 62 | 63 | 0 | 65 | 30 | 0 | 66 | 44 | 0 | 66 | 40 | 0 |
| 156 | None | T | OS | 66 | 74 | 0 | 65 | 29 | 0 | 63 | 44 | 0 | 66 | 31 | 0 |
| 157 | None | T | OS | 60 | 74 | 0 | 60 | 24 | 0 | 60 | 46 | 0 | 56 | 44 | 0 |
| 158 | None | T | OS | 50 | 67 | 10 | 53 | 30 | 10 | 53 | 50 | 10 | 52 | 45 | 10 |

§§ Beam failed but left under exposure.

(Sheet 7)

* Damaged when beam 147 failed, but left under exposure.

* Some pulse velocity readings obtained at 149 and 150 are not believed to be valid due to the power limitations of the test equipment; these ΔV° readings are therefore put in italics.

A pulse velocity reading was not obtained on this specimen.

(Revised August 1977)

Table 1-TC-B (Continued)

Section 1

Beach Row

| Beam No. | Nominal Stress psi | Steel Position | Type Steel Deformation | 2624 Cycles, 1973 | | 2760 Cycles, 1974 | | 1973-1976 Readings | | | | 3018 Cycles, 1977 | |
|----------|--------------------|----------------|------------------------|-------------------|------------------------|-------------------|------------------------|--------------------|------------------------|--------|------------------------|-------------------|------------------------|
| | | | | Con- | Max Crack Width 1/1000 | Con- | Max Crack Width 1/1000 | Con- | Max Crack Width 1/1000 | Con- | Max Crack Width 1/1000 | Con- | Max Crack Width 1/1000 |
| | | | | dition | $\%V^2$ | dition | $\%V^2$ | dition | $\%V^2$ | dition | $\%V^2$ | dition | $\%V^2$ |
| 83 | 20,000 | B | A-305 | 59 | ** | 10 | 53 | 5 | 59 | 74 | 10 | 46 | 66 |
| 84 | 20,000 | B | A-305 | 63 | | 10 | 67 | 10 | 65 | 67 | 15 | 64 | 74 |
| 85 | 20,000 | B | OS | 72 | | 25 | 71 | 20 | 71 | 72 | 25 | 66 | 73 |
| 86 | 20,000 | B | OS | 66 | | 20 | 66 | 20 | 66 | 66 | 20 | 63 | 65 |
| 87 | 20,000 | B | A-305 | 46 | | 15 | 48 | 15 | 45 | 70 | 20 | 47 | 105 |
| 88 | 20,000 | B | A-305 | 50 | | 10 | 49 | 15 | 49 | 68 | 20 | 51 | 69 |
| 89 | 20,000 | B | OS | 64 | | 20 | 66 | 20 | 65 | 76 | 25 | 64 | 76 |
| 90 | 20,000 | B | OS | 61 | | 20 | 60 | 20 | 60 | 60 | 30 | 57 | 97 |
| 91 | 30,000 | B | A-305 | 69 | | 20 | 68 | 20 | 67 | 56 | 30 | 68 | 55 |
| 92 | 30,000 | B | A-305 | 67 | | 15 | 66 | 15 | 66 | 83 | 20 | 51 | 80 |
| 93 | 30,000 | B | OS | 64 | | 25 | 66 | 40 | 64 | 53 | 35 | 64 | 53 |
| 94 | 30,000 | B | OS | 67 | | 55 | 67 | 70 | 64 | 57 | 70 | 61 | 65 |
| 95 | 30,000 | B | A-305 | 68 | | 20 | 68 | 15 | 67 | 78 | 25 | 62 | 77 |
| 96 | 30,000 | B | A-305 | 65 | | 20 | 66 | 20 | 67 | 3 | 25 | 62 | 80 |
| 97 | 30,000 | B | OS | 63 | | 40 | 63 | 30 | 61 | 59 | 50 | 62 | 58 |
| 98 | 30,000 | B | OS | 59 | | 50 | 58 | 50 | 60 | 63 | 50 | 58 | 62 |
| 99 | 40,000 | B | A-305 | 59 | | 50 | 59 | 60 | 59 | 57 | 50 | 57 | 58 |
| 100 | 40,000 | B | A-305 | 58 | | 40 | 56 | 30 | 57 | 73 | 50 | 53 | 71 |
| 101 | 40,000 | B | OS | 58 | | 60 | 60 | 70 | 60 | 53 | 75 | 58 | 52 |
| 102 | 40,000 | B | OS | 61 | | 60 | 63 | 60 | 63 | 46 | 70 | 59 | 49 |
| 103 | 40,000 | B | A-305 | 50 | | 60 | 51 | 60 | 48 | 89 | 60 | 47 | 94 |
| 104 | 40,000 | B | A-305 | 57 | | 60 | 58 | 50 | 58 | 76 | 55 | 57 | 81 |
| 105 | 40,000 | B | OS | 69 | | 50 | 70 | 50 | 68 | 110 | 55 | 64 | 108 |
| 106 | 40,000 | B | OS | 52 | | 50 | 52 | 50 | 52 | 63 | 70 | 48 | 66 |
| 107 | 50,000 | B | A-305 | 51 | | 50 | 51 | 60 | 51 | 38 | 55 | 51 | 38 |
| 108 | 50,000 | B | A-305 | 55 | | 50 | 52 | 50 | 54 | 37 | 55 | 57 | 38 |
| 109 | 50,000 | B | OS | 66 | | 60 | 66 | 70 | 66 | 62 | 60 | 64 | 126 |
| 110 | 50,000 | B | OS | 55 | | 60 | 55 | 60 | 53 | 57 | 60 | 55 | 100 |
| 111 | 50,000 | B | A-305 | 51 | | 40 | 50 | 50 | 51 | 83 | 50 | 49 | 92 |
| 112 | 50,000 | B | A-305 | 63 | | 60 | 62 | 70 | 58 | 74 | 60 | 61 | 103 |
| 113 | 50,000 | B | OS | 48 | | 80 | 48 | 80 | 48 | 75 | 80 | 47 | 76 |
| 114 | 50,000 | B | OS | 51 | | 70 | 52 | 60 | 52 | 71 | 75 | 51 | 138 |
| 115 | None | B | A-305 | 63 | | 0 | 57 | 0 | 56 | 79 | | 47 | 79 |
| 116 | None | B | A-305 | 65 | | 0 | 58 | 0 | 57 | 82 | | 55 | 79 |
| 117 | None | B | A-305 | 43 | | 0 | 45 | 0 | 48 | 42 | | 35 | 42 |
| 118 | None | B | OS | 34 | | 0 | 35 | 0 | 38 | 61 | | 35 | 65 |
| 119 | None | B | OS | 56 | | 0 | 49 | 0 | 49 | 40 | | 50 | 43 |
| 120 | None | B | OS | 53 | | 0 | 54 | 0 | 48 | 83 | | 52 | 80 |
| 121 | 20,000 | T | A-305 | 74 | | 30 | 72 | 25 | 72 | 97 | 35 | 72 | 98 |
| 122 | 20,000 | T | A-305 | 69 | | 20 | 68 | 15 | 67 | 90 | 25 | 69 | 102 |
| 123 | 20,000 | T | OS | 60 | | 50 | 61 | 50 | 60 | 90 | 50 | 60 | 103 |
| 124 | 20,000 | T | OS | 64 | | 40 | 64 | 40 | 62 | 86 | 50 | 61 | 81 |
| 125 | 20,000 | T | A-305 | 60 | | 25 | 59 | 20 | 56 | 55 | 30 | 57 | 56 |
| 126 | 20,000 | T | A-305 | 76 | | 25 | 77 | 25 | 74 | 57 | 30 | 75 | 58 |
| 127 | 20,000 | T | OS | 65 | | 15 | 67 | 20 | 66 | 57 | 20 | 67 | 56 |
| 128 | 20,000 | T | OS | 58 | | 10 | 67 | 10 | 64 | 61 | 20 | 68 | 59 |
| 129 | 30,000 | T | A-305 | 54 | | 50 | 52 | 40 | 54 | 75 | 50 | 52 | 98 |
| 130 | 30,000 | T | A-305 | 63 | | 50 | 63 | 50 | 63 | 70 | 60 | 61 | 111 |
| 131 | 30,000 | T | OS | 72 | | 60 | 72 | 60 | 70 | 62 | 70 | 72 | 62 |
| 132 | 30,000 | T | OS | 59 | | 60 | 59 | 60 | 58 | 71 | 60 | 51 | 83 |
| 133 | 30,000 | T | A-305 | 60 | | 50 | 61 | 50 | 58 | 83 | 50 | 57 | 78 |
| 134 | 30,000 | T | A-305 | 64 | | 60 | 64 | 50 | 64 | 60 | 65 | 62 | 61 |
| 135 | 30,000 | T | OS | 61 | | 40 | 61 | 40 | 61 | 50 | 40 | 61 | 46 |
| 136 | 30,000 | T | OS | 65 | | 20 | 65 | 30 | 62 | 82 | 30 | 63 | 86 |
| 137 | 40,000 | T | A-305 | 49 | | 80 | 50 | 70 | 49 | 66 | 85 | 49 | 67 |
| 138 | 40,000 | T | A-305 | 56 | | 75 | 55 | 80 | 56 | 68 | 75 | 55 | 67 |
| 139 | 40,000 | T | OS | 66 | | 50 | 65 | 60 | 65 | 92 | 50 | 64 | 68 |
| 140 | 40,000 | T | OS | 67 | | 45 | 67 | 40 | 67 | 65 | 55 | 66 | 67 |
| 141 | 40,000 | T | A-305 | 53 | | 60 | 53 | 60 | 52 | 75 | 60 | 58 | 105 |
| 142 | 40,000 | T | A-305 | 53 | | 40 | 59 | 50 | 54 | 70 | 50 | 61 | 86 |
| 143 | 40,000 | T | OS | 64 | | 40 | 63 | 40 | 62 | 78 | 50 | 64 | 114 |
| 144 | 40,000 | T | OS | 61 | | 60 | 61 | 70 | 55 | 72 | 65 | 59 | 91 |
| 145 | 50,000 | T | A-305 | 60 | | 80 | 54 | 80 | 52 | 68 | 80 | 54 | 116 |
| 146 | 50,000 | T | A-305 | 46 | | 50 | 48 | 60 | 44 | 72 | 70 | 46 | 84 |

** Satisfactory pulse velocity readings were not obtained in 1973 and 1974.

(Sheet 8)

(Revised August 1980)

Table 1-TC-B (Continued)

Section 2

| Beam No. | Nominal Stress, ksi | Steel Position | Type Steel Deformation | Con- di- tion | 3095 Cycles, 1977 | | 1977- Headings 3242 Cycles, 1978 | | 3335 Cycles, 1977 | |
|----------|---------------------|----------------|------------------------|---------------------|--|---------------------|--|---------------------|--|---------------------|
| | | | | | Max Crack Width 1/1000 in. | Con- di- tion | Max Crack Width 1/1000 in. | Con- di- tion | Max Crack Width 1/1000 in. | Con- di- tion |
| | | | | | | | | | | |
| | | | | | | | | | | |
| 84 | 20,000 | B | A-305 | 22 | 80 | 15 | 47 | 78 | 15 | 85 |
| 84 | 20,000 | B | A-305 | 60 | 74 | 25 | 64 | 47 | 30 | 64 |
| 85 | 20,000 | B | OS | 68 | 88 | 27 | 66 | 40 | 20 | 59 |
| 86 | 20,000 | B | OS | 65 | 69 | 25 | 67 | 58 | 25 | 63 |
| 87 | 20,000 | B | A-305 | 46 | 66 | 20 | 46 | 52 | 20 | 44 |
| 88 | 20,000 | B | A-305 | 51 | 53 | 20 | 53 | 52 | 20 | 46 |
| 89 | 20,000 | B | OS | 63 | 46 | 25 | 64 | 56 | 25 | 61 |
| 90 | 20,000 | B | OS | 58 | 59 | 25 | 56 | 49 | 20 | 58 |
| 91 | 30,000 | B | A-305 | 68 | 60 | 25 | 69 | 38 | 30 | 68 |
| 92 | 30,000 | B | A-305 | 65 | 84 | 25 | 66 | 39 | 25 | 64 |
| 93 | 30,000 | B | OS | 63 | 51 | 55 | 65 | 44 | 60 | 64 |
| 94 | 30,000 | B | OS | 63 | 68 | 70 | 65 | 60 | 75 | 60 |
| 95 | 30,000 | B | A-305 | 60 | 76 | 25 | 46 | 63 | 20 | 42 |
| 96 | 30,000 | B | A-305 | 64 | 86 | 25 | 65 | 60 | 96 | 63 |
| 97 | 30,000 | B | OS | 62 | 76 | 50 | 63 | 66 | 40 | 62 |
| 98 | 30,000 | B | OS | 59 | 63 | 50 | 61 | 54 | 50 | 55 |
| 99 | 40,000 | B | A-305 | 56 | 92 | 60 | 59 | 54 | 75 | 59 |
| 100 | 40,000 | B | A-305 | 56 | 72 | 55 | 56 | 41 | 50 | 43 |
| 101 | 40,000 | B | OS | 56 | 54 | 80 | 57 | 43 | 80 | 56 |
| 102 | 40,000 | B | OS | 58 | 53 | 100 | 56 | 58 | 125 | 54 |
| 103 | 40,000 | B | A-305 | 47 | 86 | 60 | 47 | 57 | 60 | 46 |
| 104 | 40,000 | B | A-305 | 64 | 62 | 60 | 58 | 73 | 75 | 56 |
| 105 | 40,000 | B | OS | 65 | 82 | 75 | 65 | 88 | 80 | 88 |
| 106 | 40,000 | B | OS | 66 | 69 | 70 | 90 | 76 | (1/4-in. spall) | 91 |
| 107 | 50,000 | B | A-305 | 50 | 46 | (1-in. spall) | 51 | 55 | (1-in. spall) | 49 |
| 108 | 50,000 | B | A-305 | 53 | 44 | (5/8-in. spall) | 54 | 33 | (5/8-in. spall) | 49 |
| 109 | 50,000 | B | OS | 65 | 115 | (1-1/2 in. spall) | 64 | 54 | (2-in. spall) | 56 |
| 110 | 50,000 | B | OS | 53 | 94 | 75 | 55 | 62 | 100 | 53 |
| 111 | 50,000 | B | A-305 | 50 | 68 | 75 | 55 | 43 | 100 | 49 |
| 112 | 50,000 | B | A-305 | 59 | 67 | 75 | 61 | 50 | (1-in. spall) | 62 |
| 113 | 50,000 | B | OS | 47 | 79 | (1/4-in. spall) | 48 | 48 | (1/4-in. spall) | 46 |
| 114 | 50,000 | B | OS | 51 | 101 | 100 | 51 | 56 | (1/2-in. spall) | 51 |
| 115 | None | B | A-305 | 51 | 78 | | 49 | 43 | | 46 |
| 116 | None | B | A-305 | 57 | 76 | | 57 | 25 | | 62 |
| 117 | None | B | A-305 | 59 | 44 | | 53 | 45 | 30 | 52 |
| 118 | None | B | OS | 44 | 64 | | 42 | 22 | | 39 |
| 119 | None | B | OS | 55 | 43 | | 52 | 70 | | 45 |
| 120 | None | B | OS | 54 | 79 | | 55 | 50 | | 53 |
| 121 | 20,000 | T | A-305 | 70 | 92 | 35 | 73 | 52 | 30 | 71 |
| 122 | 20,000 | T | A-305 | 67 | 97 | 35 | 69 | 60 | 30 | 67 |
| 123 | 20,000 | T | OS | 59 | 65 | 60 | 59 | 51 | 100 | 52 |
| 124 | 20,000 | T | OS | 63 | 71 | 60 | 63 | 47 | 60 | 63 |
| 125 | 20,000 | T | A-305 | 57 | 54 | 30 | 59 | 30 | 25 | 57 |
| 126 | 20,000 | T | A-305 | 75 | 54 | 30 | 76 | 37 | 25 | 74 |
| 127 | 20,000 | T | OS | 66 | 68 | 20 | 68 | 61 | 20 | 55 |
| 128 | 20,000 | T | OS | 67 | 65 | 20 | 68 | 35 | 15 | 57 |
| 129 | 30,000 | T | A-305 | 52 | 93 | 50 | 54 | 96 | 60 | 53 |
| 130 | 30,000 | T | A-305 | 60 | 83 | 60 | 61 | 69 | 75 | 56 |
| 131 | 30,000 | T | OS | 70 | 49 | 75 | 72 | 63 | 75 | 72 |
| 132 | 30,000 | T | OS | 54 | 53 | 75 | 56 | 69 | (1/4-in. spall) | 47 |
| 133 | 30,000 | T | A-305 | 58 | 91 | 50 | 58 | 47 | 50 | 58 |
| 134 | 30,000 | T | A-305 | 62 | 70 | 75 | 64 | 54 | 50 | 63 |
| 135 | 30,000 | T | OS | 61 | 43 | 40 | 61 | 29 | 40 | 63 |
| 136 | 30,000 | T | OS | 66 | 59 | 40 | 63 | 48 | 50 | 64 |
| 137 | 30,000 | T | A-305 | 49 | 79 | 100 | 50 | 57 | 110 | 44 |
| 138 | 30,000 | T | A-305 | 60 | 77 | 75 | 58 | 46 | 75 | 58 |
| 139 | 30,000 | T | OS | 60 | 80 | 75 | 66 | 49 | 75 | 64 |
| 140 | 30,000 | T | OS | 60 | 81 | 75 | 67 | 49 | 75 | 64 |
| 141 | 30,000 | T | A-305 | 59 | 94 | 75 | 53 | 75 | 75 | 56 |
| 142 | 30,000 | T | A-305 | 57 | 86 | 60 | 60 | 75 | 75 | 54 |
| 143 | 30,000 | T | OS | 61 | 81 | 75 | 63 | 71 | 75 | 61 |
| 144 | 30,000 | T | OS | 60 | 93 | (1/2-in. spall) | 64 | 71 | (1/2-in. spall) | 61 |
| 145 | 30,000 | T | A-305 | 61 | 89 | 100 | 60 | 71 | 75 | 61 |
| 146 | 30,000 | T | A-305 | 67 | 73 | 75 | 67 | 71 | 75 | 61 |

(Continued)

(Revised July 1981)

Table 1-TC-B (Continued)

Section 2

Beach Row 1

| Beam No. | Nominal Stress psi | Steel Position | Type Steel Deformation | 1973-1976 Readings | | | | | | | | | |
|----------|--------------------|----------------|------------------------|--------------------|--|-------------------|--|-------------------|--|-------------------|--|----------------|--|
| | | | | 2654 Cycles, 1973 | | 2760 Cycles, 1974 | | 2872 Cycles, 1975 | | 3018 Cycles, 1976 | | | |
| | | | | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. |
| 148 | 50,000 | T | OS | ## | 75 | Unloaded | ## | -- | -- | -- | -- | -- | -- |
| 149 | 50,000 | T | A-305 | 68 | 66 | 500# | 61 | 73 | 500 | 61 | 100 | (4-in. spall) | 75 |
| 150 | 50,000 | T | A-305 | 65 | 60 | 65 | 70 | 62 | 61 | 70 | 64 | 104 | 75 |
| 151 | 50,000 | T | OS | 57 | 70 | 58 | 70 | 57 | 62 | 60 | 56 | 63 | (1/2-in. spall) |
| 152 | 50,000 | T | OS | 54 | 60 | 54 | 55 | 51 | 52 | 50 | 53 | 52 | 50 |
| 153 | None | T | A-305 | 44 | 0 | 36 | 0 | 16 | 50 | | 26 | 52 | |
| 154 | None | T | A-305 | 55 | 0 | 54 | 0 | 54 | 84 | | 55 | 81 | |
| 155 | None | T | A-305 | 76 | 0 | 65 | 0 | 61 | 82 | | 65 | 81 | |
| 156 | None | T | OS | 52 | 0 | 27 | 0 | 25 | 83 | | 22 | 81 | |
| 157 | None | T | OS | 52 | 0 | 51 | 0 | 49 | 83 | | 50 | 90 | |
| 158 | None | T | OS | 51 | 0 | 50 | 0 | 50 | 74 | | 51 | 79 | (2-in. spall) |

| Beam No. | Nominal Stress psi | Steel Position | Type Steel Deformation | 1977- Readings | | | | | | | | | |
|----------|--------------------|----------------|------------------------|-------------------|--|-------------------|--|-------------------|--|----------------|--|----------------|--|
| | | | | 3095 Cycles, 1977 | | 3242 Cycles, 1978 | | 3341 Cycles, 1979 | | | | | |
| | | | | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. | Con- dition | Max Crack Width 1/1000 in. |
| 148 | 50,000 | T | OS | -- | -- | -- | -- | -- | -- | -- | -- | -- | -- |
| 149 | 50,000 | T | A-305 | 64 | 67 (4-in. spall) | 62 | 50 | (6-in. spall) | 61 | 93 | (6-in. spall) | | |
| 150 | 50,000 | T | A-305 | 63 | 65 | 63 | 60 | 75 | 62 | 90 | 75 | | |
| 151 | 50,000 | T | OS | 53 | 63 (5/8-in. spall) | 45 | 32 | (1-in. spall) | 45 | 54 | (1-in. spall) | | |
| 152 | 50,000 | T | OS | 53 | 67 | 54 | 33 | 50 | 52 | 96 | 50 | | |
| 153 | None | T | A-305 | 29 | 53 | 26 | 35 | | 19 | 55 | -- | | |
| 154 | None | T | A-305 | 55 | 74 | 61 | 24 | | 59 | -- | -- | | |
| 155 | None | T | A-305 | 65 | 82 | 66 | 53 | | 65 | 52 | -- | | |
| 156 | None | T | OS | 23 | 82 | 23 | 52 | | 21 | 56 | -- | | |
| 157 | None | T | OS | 54 | 80 | 57 | 28 | | 49 | 82 | -- | | |
| 158 | None | T | OS | 50 | 68 | 52 | 36 | (2-in. spall) | 50 | 74 | (2-in. spall) | | |

Note: Fatigue of these beams was discontinued after 1979.

Satisfactory pulse velocity readings were not obtained in 1973 and 1974.

* One rebar failed during winter of 1974-1975.

APPENDIX B: ANALYSIS OF VARIANCE

TENSILE CRACK, SERIES 9, REINFORCED CONCRETE
LONG-TERM DURABILITY TEST
1957-1979
ANALYSIS OF VARIANCE

ANALYSIS OF VARIANCE PROCEDURE

DEPENDENT VARIABLE: COND

CONDITION

| SOURCE | DF | SUM OF SQUARES | MEAN SQUARE | F VALUE | PR > F | R-SQUARE | C.V. |
|-----------------|-----|----------------|--------------|---------|------------|----------|-------------|
| MODEL | 107 | 46074.27524240 | 152.68591303 | 10.78 | 0.0001 | 0.978714 | 5.9220 |
| ERROR | 72 | 1014.47324501 | 14.15935065 | | STD DEV | | COAC MEAN |
| CORRECTED TOTAL | 379 | 47094.04053801 | | | 3.76289126 | | 63.54122807 |

B2

| SOURCE | DF | ANUVA SS | F VALUE | PR > F |
|------------------------|----|----------------|---------|--------|
| PLSIT | 1 | 310.27368421 | 9.35 | 0.0051 |
| TYPE | 1 | 363.5100187 | 25.57 | 0.0001 |
| PLSIT*TYPE | 1 | 90.07397661 | 6.83 | 0.0109 |
| STRESS | 4 | 4313.21739766 | 146.78 | 0.0001 |
| PLSIT*STRESS | 4 | 1101.40394737 | 19.45 | 0.0001 |
| TYPE*STRESS | 4 | 3074.57441520 | 64.89 | 0.0001 |
| PLSIT*TYPE*STRESS | 4 | 2719.256097661 | 48.02 | 0.0001 |
| YEAR | 14 | 24744.05367690 | 97.09 | 0.0001 |
| PLSIT*YEAR | 14 | 295.087970801 | 1.12 | 0.3504 |
| TYPE*YEAR | 14 | 53.491794590 | 0.81 | 0.9997 |
| PLSIT*TYPE*YEAR | 14 | 3035.76593567 | 2.03 | 0.0119 |
| PLSIT*TYPE*STRESS*YEAR | 14 | 758.94010228 | 2.98 | 0.0055 |
| PLSIT*STRESS*YEAR | 72 | 1750.43910819 | 1.72 | 0.0110 |
| TYPE*STRESS*YEAR | 72 | 657.94919491 | 0.84 | 0.7670 |

TENSILE CRACK, SERIES B REINFORCED CONCRETE
LONG-TERM DURABILITY TEST
1957-1979
ANALYSIS OF VARIANCE

ANALYSIS OF VARIANCE PROCEDURE

DEPENDENT VARIABLE: PCT_V2 PERCENT VELOCITY SQUARED

| SOURCE | DF | SUM OF SQUARES | MEAN SQUARE | F VALUE | PR > F | R-SQUARE | C.V. |
|-----------------|-----|-----------------|--------------|---------|------------|----------|-------------|
| MODEL | 307 | 320142.32062233 | 1045.9899276 | 33.51 | 0.0001 | 0.993050 | 9.2165 |
| ERROR | 72 | 2245.42290232 | 31.18647920 | | STD DEV | | PCT_V2 MEAN |
| CORRECTED TOTAL | 379 | 323037.74352465 | | | 5.58448110 | | 60.59252754 |

| SOURCE | DF | ANUVA SS | F VALUE | PR > F |
|-------------------------|----|-----------------|---------|--------|
| POSIT | 1 | 14.45264059 | 0.46 | 0.4976 |
| TYPE | 1 | 158.0393300 | 5.07 | 0.0274 |
| POSIT*TYPE | 1 | 13.72146568 | 0.44 | 0.5072 |
| SPAC*SU | 4 | 5504.5830128 | 44.16 | 0.0001 |
| POSIT*SPAC*SU | 4 | 1379.27180721 | 8.65 | 0.0001 |
| TYPE*SPAC*SU | 4 | 321.91921936 | 2.53 | 0.0443 |
| POSIT*TYPE*SPAC*SU | 4 | 1971.55722661 | 13.40 | 0.0031 |
| YEAR | 14 | 302071.13714371 | 539.11 | 0.0001 |
| POSIT*YEAR | 14 | 246.31240117 | 0.42 | 0.9779 |
| TYPE*YEAR | 14 | 237.07126304 | 0.37 | 0.9999 |
| POSIT*TYPE*YEAR | 14 | 4357.36678237 | 2.39 | 0.0071 |
| POSIT*TYPE*YEAR*SU | 13 | 546.37964539 | 0.97 | 0.6498 |
| POSIT*TYPE*YEAR*SU*SPAC | 72 | 1945.10150044 | 0.86 | 0.7317 |
| TYPE*YEAR*SU*SPAC | 72 | 1712.30683448 | 0.75 | 0.8715 |

TENSILE CRACK, SERIES 4 REINFORCED CONCRETE
LONG-TERM DURABILITY TEST
1957-1979
ANALYSIS OF VARIANCE

ANALYSIS OF VARIANCE PROCEDURE

DEPENDENT VARIABLE: MAX_C MAXIMUM CRACK WIDTH

| SOURCE | DF | SUM OF SQUARES | MEAN SQUARE | F VALUE | PR > F | R-SQUARE | C.V. |
|-----------------|-----|----------------|-------------|---------|------------|----------|------------|
| MODEL | 249 | 4.76019698 | 0.03518151 | 2.51 | 0.0001 | 0.920434 | 169.0487 |
| ERROR | 54 | 0.75726135 | 0.01402337 | | STO DEV | | MAX_C MEAN |
| CORRECTED TOTAL | 303 | 9.51745833 | | | 0.11842030 | | 0.0705099 |

B4

| SOURCE | DF | ANOVA SS | F VALUE | PR > F |
|-----------------|----|------------|---------|--------|
| PCSLT | 1 | 0.05151685 | 3.82 | 0.0559 |
| TYPE | 1 | 0.01638024 | 1.17 | 0.2846 |
| PLSLTTYPE | 1 | 0.05528757 | 3.94 | 0.0522 |
| STRESS | 3 | 1.25034329 | 29.96 | 0.0001 |
| PCSLTSTRESS | 3 | 0.07578940 | 1.80 | 0.1564 |
| TYPESTRESS | 3 | 0.12232046 | 2.91 | 0.0422 |
| PLSLTTYPESTRESS | 3 | 0.17069319 | 4.06 | 0.0114 |
| YEAR | 19 | 1.96758102 | 7.79 | 0.0001 |
| PCSLTYEAR | 18 | 0.12692738 | 0.50 | 0.9454 |
| TYPEYEAR | 13 | 0.10778392 | 0.43 | 0.9753 |
| STRESSYEAR | 54 | 3.04614699 | 5.08 | 0.0001 |
| PLSLTYEAR | 10 | 0.19226323 | 0.76 | 0.7327 |
| TYPESTRESSYEAR | 54 | 0.19000252 | 0.25 | 1.0000 |
| TYPESTRESSYEAR | 54 | 0.57220592 | 0.76 | 0.9469 |

APPENDIX C: LINEAR REGRESSION ANALYSES

Condition

| <u>Position</u> | <u>Type Rebar</u> | <u>Stress kips</u> | <u>Correlation Coefficient</u> | <u>Regression Equation</u> |
|-----------------|-------------------|--------------------|--------------------------------|-----------------------------------|
| Bottom | A-305 | 0 | -0.95 | Condition = 166.42 - 1.48 * Year |
| Bottom | A-305 | 20 | -0.94 | Condition = 164.56 - 1.51 * Year |
| Bottom | A-305 | 30 | -0.93 | Condition = 134.95 - 0.947 * Year |
| Bottom | A-305 | 40 | -0.91 | Condition = 140.37 - 1.15 * Year |
| Bottom | A-305 | 50 | -0.79 | Condition = 122.14 - 0.919 * Year |
| Bottom | OS | 0 | -0.78 | Condition = 132.39 - 1.14 * Year |
| Bottom | OS | 20 | -0.91 | Condition = 140.55 - 1.03 * Year |
| Bottom | OS | 30 | -0.83 | Condition = 122.70 - 0.82 * Year |
| Bottom | OS | 40 | -0.60 | Condition = 107.63 - 0.606 * Year |
| Bottom | OS | 50 | -0.75 | Condition = 100.47 - 0.629 * Year |
| Top | A-305 | 0 | -0.95 | Condition = 156.23 - 1.41 * Year |
| Top | A-305 | 20 | -0.83 | Condition = 117.27 - 0.65 * Year |
| Top | A-305 | 30 | -0.83 | Condition = 125.84 - 0.90 * Year |
| Top | A-305 | 40 | -0.71 | Condition = 106.36 - 0.709 * Year |
| Top | A-305 | 50 | -0.90 | Condition = 134.21 - 1.03 * Year |
| Top | OS | 0 | -0.96 | Condition = 169.81 - 1.63 * Year |
| Top | OS | 20 | -0.89 | Condition = 139.14 - 1.02 * Year |
| Top | OS | 30 | -0.88 | Condition = 113.68 - 0.691 * Year |
| Top | OS | 40 | -0.88 | Condition = 106.12 - 0.568 * Year |
| Top | OS | 50 | -0.93 | Condition = 207.22 - 2.42 * Year |

Percent v^2

| | | | | |
|--------|-------|----|-------|------------------------------------|
| Bottom | A-305 | 0 | -0.77 | PCT - v^2 = 270.94 - 3.02 * Year |
| Bottom | A-305 | 20 | -0.85 | PCT - v^2 = 299.35 - 3.50 * Year |
| Bottom | A-305 | 30 | -0.82 | PCT - v^2 = 277.33 - 3.91 * Year |
| Bottom | A-305 | 40 | -0.86 | PCT - v^2 = 312.28 - 3.75 * Year |
| Bottom | A-305 | 50 | -0.88 | PCT - v^2 = 310.77 - 3.78 * Year |
| Bottom | OS | 0 | -0.82 | PCT - v^2 = 269.71 - 3.04 * Year |
| Bottom | OS | 20 | -0.84 | PCT - v^2 = 282.08 - 3.28 * Year |
| Bottom | OS | 30 | -0.83 | PCT - v^2 = 291.73 - 3.41 * Year |
| Bottom | OS | 40 | -0.85 | PCT - v^2 = 299.03 - 3.58 * Year |
| Bottom | OS | 50 | -0.86 | PCT - v^2 = 289.65 - 3.41 * Year |
| Top | A-305 | 0 | -0.80 | PCT - v^2 = 281.15 - 3.17 * Year |
| Top | A-305 | 20 | -0.79 | PCT - v^2 = 260.93 - 2.97 * Year |
| Top | A-305 | 30 | -0.81 | PCT - v^2 = 264.87 - 3.04 * Year |
| Top | A-305 | 40 | -0.84 | PCT - v^2 = 294.72 - 3.46 * Year |
| Top | A-305 | 50 | -0.88 | PCT - v^2 = 296.11 - 3.61 * Year |

(Continued)

Percent V^2 (Continued)

| <u>Position</u> | <u>Type Rebar</u> | <u>Stress kips</u> | <u>Correlation Coefficient</u> | <u>Regression Equation</u> |
|-----------------|-------------------|--------------------|--------------------------------|---|
| Top | OS | 0 | -0.71 | PCT - $V^2 = 246.35 - 2.64 * \text{Year}$ |
| Top | OS | 20 | -0.86 | PCT - $V^2 = 304.80 - 3.60 * \text{Year}$ |
| Top | OS | 30 | -0.82 | PCT - $V^2 = 268.19 - 3.11 * \text{Year}$ |
| Top | OS | 40 | -0.83 | PCT - $V^2 = 299.00 - 3.47 * \text{Year}$ |
| Top | OS | 50 | -0.91 | PCT - $V^2 = 375.29 - 4.88 * \text{Year}$ |

Maximum Crack Width

| | | | | |
|--------|-------|----|------|---|
| Bottom | A-305 | 20 | 0.80 | Max crack width = $-0.0227 + 0.000565 * \text{Year}$ |
| Bottom | A-305 | 30 | 0.70 | Max crack width = $-0.0336 + 0.000827 * \text{Year}$ |
| Bottom | A-305 | 40 | 0.95 | Max crack width = $-0.1297 + 0.00247 * \text{Year}$ |
| Bottom | A-305 | 50 | 0.72 | Max crack width = $-1.28 + 0.0208 * \text{Year}$ |
| Bottom | OS | 20 | 0.80 | Max crack width = $-0.0299 + 0.000711 * \text{Year}$ |
| Bottom | OS | 30 | 0.96 | Max crack width = $-0.1084 + 0.00213 * \text{Year}$ |
| Bottom | OS | 40 | 0.97 | Max crack width = $-0.167 + 0.00315 * \text{Year}$ |
| Bottom | OS | 50 | 0.75 | Max crack width = $-1.47 + 0.024 * \text{Year}$ |
| Top | A-305 | 20 | 0.75 | Max crack width = $-0.0370 + 0.000853 * \text{Year}$ |
| Top | A-305 | 30 | 0.93 | Max crack width = $-0.129 + 0.00236 * \text{Year}$ |
| Top | A-305 | 40 | 0.96 | Max crack width = $-0.1562 + 0.00303 * \text{Year}$ |
| Top | A-305 | 50 | 0.75 | Max crack width = $-3.53 + 0.0570 * \text{Year}$ |
| Top | OS | 20 | 0.91 | Max crack width = $-0.0950 + 0.00177 * \text{Year}$ |
| Top | OS | 30 | 0.93 | Max crack width = $-0.0862 + 0.00178 * \text{Year}$ |
| Top | OS | 40 | 0.70 | Max crack width = $-0.504 + 0.00845 * \text{Year}$ |
| Top | OS | 50 | 0.61 | Max crack width = $-0.766 + 0.013 * \text{Year}$ |

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