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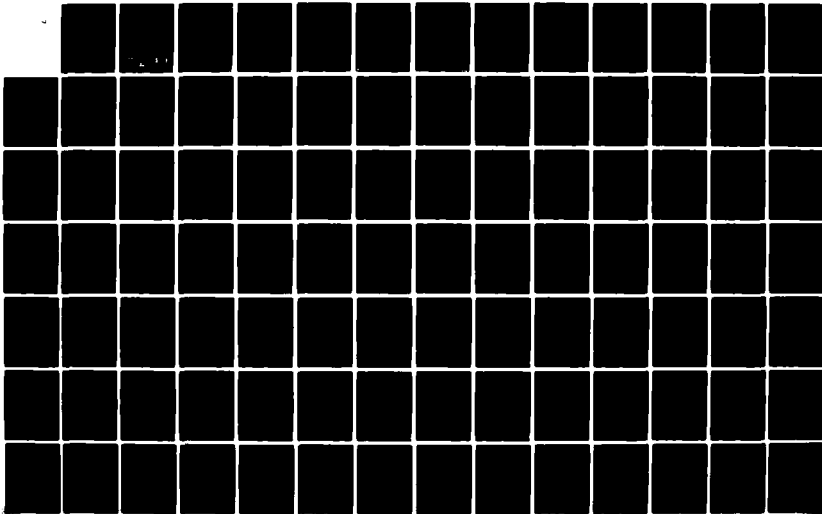
ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR  
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ADMINISTRATION GREENBELT MD GO. . E G STASSINOPOULOS  
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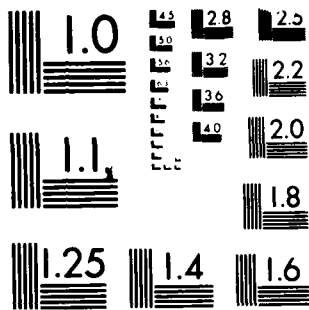
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# ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR TRAJECTORIES

Contract N00173-81-MP-06630

E.G. STASSINOPOULOS

NOVEMBER 1981

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## NASA

National Aeronautics and  
Space Administration

Goddard Space Flight Center,  
Greenbelt, Maryland 20771

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Orbital Radiation Study for Inclined  
Circular Trajectories

E. G. Stassinopoulos

NASA-Goddard Space Flight Center  
Sciences Directorate  
National Space Science Data Center



November 1981

Goddard Space Flight Center  
Greenbelt, Maryland 20771

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## 1. Introduction

At the request of and with support from the Naval Research Laboratory\*, a comprehensive study was conducted to determine the space radiation environment of a series of high inclination geocentric orbits for a mission duration of five years.)

Following the precedent established with ~~previous studies~~, the external (surface incident) charged particle radiation, predicted for the satellite was determined by orbital flux integration for six independent trajectories, (see section 3). The latest standard models of the environment were used in the calculations, (see section 5).

Magnetic field definitions for the six nominal circular trajectories were obtained from a current field model, (see section 3).

Spatial and temporal variations or conditions affecting the static environment models were considered and accounted for, wherever possible.

Finally, limited shielding and dose evaluations were performed for simple infinite slab and spherical geometries.

Results, given in graphical and tabular form, are analyzed, explained, and discussed. Conclusions are presented and commented on.

## 2. SPECIFICATION OF ORBITS

The analysis was based on nominal circular orbits with inclination of 60 degrees and altitudes of 1667, 2593, 3889, 5186, 6389, and 10371 kilometers.

## 3. GENERATION OF TRAJECTORIES

Six separate flight path ephemerides were generated for the specified orbits with the GEODYN-BLCONV System<sup>1</sup> for trajectories of 24-hour duration defined at 1-minute intervals. The length of simulated orbit time and the integration stepsize were especially selected so as to provide sufficient point density to insure an adequate sampling of the ambient radiation environment when flying the trajectories through the models. The trajectories were subsequently converted from geodetic polar to magnetic B-L coordinates with McIlwain's INVAR program of 1965<sup>2</sup> and the field routine ALLMAG,<sup>3</sup> which now utilizes the BARRACLOUGH 1975 field model.<sup>4</sup> The field computations were extrapolated to the tentative mission epoch of 1989.5 with linear time terms representing secular variations of the field.

## 4. FLIGHT PATH EXPOSURE TO TRAPPING DOMAINS

The specified nominal flight-path configurations display a significant characteristic of high inclination orbits in magnetic L-space: they traverse almost the entire terrestrial radiation belt twice during each revolution, moving back and forth through regions of low L values (the inner zone:  $1.0 < L \leq 2.8$ ), regions of high L values (the outer zone:  $2.8 < L < 12$ ), and

\*This work funded by the Naval Electronics Systems Command under the NRL Nuclear Survivability/Vulnerability Program.

regions outside the trapping domain (external). Occasionally, some revolutions will also enter regions of space where no particle trapping can occur because of atmospheric cut-off conditions; that is, trajectory segments may have a combination of magnetic B and L values that place them outside the atmospheric cut-off limits of the models.

These excursions and the "external" visitations afford the satellite an amount of flux-free time, which may be of substantial duration (see section 11, C).

#### 5. TRAPPED PARTICLE ENVIRONMENT MODELS

The fluxes in this study were obtained from current NSSDC models: the solar maximum AE6 for the inner zone electrons<sup>5</sup>, the new interim model AEI7 for the outer zone electrons<sup>6</sup>, and the solar maximum version of the new AP8 model<sup>7</sup> for energetic trapped protons. It should be noted that the interim AEI7 does not reflect solar cycle variations in its present state. However, this model was issued in two versions, the AEI7-HI and the AEI7-LO, in order to account for differences in the data sets used in their construction. The LO version was used in this effort. All models describe an average static environment at a given epoch.

#### 6. ORBITAL FLUX INTEGRATIONS

Orbital flux integrations were performed with the UNIFLUX<sup>8</sup> and the SOFIP<sup>9</sup> systems. UNIFLUX provides L-band distributions and exposure times with B-L bin breakdown, while SOFIP provides the dose and shield data.

#### 7. GEOMAGNETIC SHIELDING AND SOLAR FLARE PROTONS

Low altitude high inclination orbits experience a significant amount of geomagnetic shielding from cosmic rays of solar or galactic origin in the energy range  $E > 10$  MeV. Therefore, it may be assumed that the spacecraft will only intermittently be exposed to the unattenuated interplanetary solar flare proton intensities of all energies above 10 MeV. To a first approximation, the fluxes may also be considered omnidirectional and isotropic, probably to within 10-15%.

Usually, geomagnetic shielding effects on geocentric missions are being evaluated with simple rigidity considerations because of substantial diurnal variations in the cutoff latitude associated with geomagnetic tail effects (2-4 degrees) and storm-induced changes (> 4 degrees). The simple analysis used here assumed that energetic solar protons of all energies above 10 MeV have free access to all magnetospheric regions external to a dipole shell of L=5 earth radii, which is equivalent to a cut-off latitude of about 63 degrees.

Predictions of solar flare proton fluxes at 1 AU are obtained as a function of mission duration  $\tau$  and confidence level  $Q^*$  on the basis of a probabilistic analysis<sup>10</sup> using a modified type of Poisson statistics by a computerized model SOLPRO<sup>11</sup> that includes the distinction between "ordinary" (OR) and

---

\* $Q$  denotes the degree of confidence one wishes to assign to the results, namely that for the specified mission duration the calculated fluences are the smallest values which will not be exceeded by actually encountered intensities.



"anomalously large" (AL) events and the probability of occurrence of the latter. Both AL- and OR- event fluences are non-linear functions of  $Q$  and  $\tau$ . For these predictions, only high quality comprehensive satellite measurements (not ground observations) are being used, covering almost the entire 20th solar cycle. There have been indications that descriptions of the solar flare environment in interplanetary space (at 1 AU), derived from interpretations and extrapolations of ground based measurements, have not been very accurate.

It should be noted that the statistics cannot predict when an AL event will occur; only the probability that one will occur in a given length of time. And it must be remembered that a single AL event will impart its total fluence within two to four days.

This implies that for unmanned satellites with mission durations of  $\tau \geq 1$  year, OR-event fluences are not significant because probabilistic theory predicts the possible occurrence of at least one AL event, even for the lowest allowable confidence level ( $Q=80\%$ ).

### 8. FLUX DATA: TYPE, QUALITY, AND VARIATIONS

The trapped particle flux data available from the models represent omnidirectional, integral intensities that one would expect to obtain as average values over periods in excess of six months. But over most regions of magnetospheric space ( $L \geq 2$  earth radii), short term excursions can vary from these values by factors of  $10^2$  to  $10^3$ , depending on the particle energies and on the type and intensity of the causative event. These variations do affect the investigated missions because their trajectories enter regions of space where  $L$  is greater than 2 earth radii. Also, trapped particle populations experience changes due to: (a) local time (LT) dependence, and (b) solar cycle dependence. Both are of some consequence to these missions. The former is significant for spacecraft that sample regions of  $L > 5$  earth radii, which are visited by the specified missions. To compensate for these variations, the model provides LT-averaged values, which should yield an adequate approximation for missions of long duration ( $\tau \geq 1$  year). The solar cycle variations have been taken into account by selecting the appropriate models for each period, where available.

Generally, solar cycle variations have opposite effects on each particle species:

	<u>Solar Min</u>	<u>Solar Max</u>
Electrons	lower	higher
Protons	higher	lower

The solar cycle changes, as derived from a comparison of the corresponding models, are functions of energy  $E$  and magnetic parameter  $L$ . For the inner zone electrons, they may range from a factor of 1 to a factor of 5.

Protons are only affected in the vicinity of the atmospheric cutoff regions. No changes of consequence have been observed in the heart of the proton trapping domain. Proton changes have about the same range as those of the electrons.

It is necessary to emphasize that the calculations, although based on the best data available for the past epochs, can only serve as approximations for the future.

It also should be noted that a basic uncertainty factor of 2 is defined for the flux values of the AP8 and the AE5 models, while the AE6 is characterized by an average uncertainty factor of 5. No uncertainty factor has yet been defined for the interim AEI7.

#### 9. DOSE AND SHIELDING EVALUATION

Doses were calculated from the total orbit integrated, surface incident, omnidirectional, integral fluences by existing shielding codes<sup>12</sup>, as functions of various aluminum shield thicknesses and geometries.

A simple procedure was followed, not involving solid angle sectoring or three-dimensional ray tracing considerations. Instead, a simple two-dimensional geometry with a cosine law for the incident spectra, and a three-dimensional spherical geometry were considered. (See comment in section 11D-III)

Bremsstrahlung calculations were performed with the same codes.

#### 10. RESULTS: PRESENTATION DESCRIPTION

This section describes the form and format in which the results, derived from the Orbital Flux Integration (OFI) process, are presented for practical use. Except where otherwise specified, all particle data in this report relate to integral, omnidirectional fluxes or fluences.

##### A. Tabular Presentations

The outcome of all calculations is summarized in Tables 1 to 60. The tables are arranged in six sets, where every set pertains to one specific type of data. The first two sets have two similar members for every trajectory considered in the study: one for trapped protons and one for electrons, in that order. The next three sets contain only one member for each trajectory. The sixth set contains three similar members for each trajectory. A more detailed description of the tables is provided in the following paragraphs.

##### I. L-band Tabulations: Tables 1-12

Tabulation of total orbit-integrated fluence distributions by L-bands for selected energy thresholds, in units of particles per square centimeter, normalized to 5 MeV and .5 MeV for protons and electrons, respectively.

The tables contain 48 L-bands of equal size covering the range from L=1.0 to L > 10.4 earth radii in constant increments of .2 earth radii.

##### II. Spectral Profiles: Tables 13-24

Tabulation of average orbit-integrated spectral distributions. Composite spectra are given in units of: fluxes per square centimeter per second, fluxes per square centimeter per day, and total fluences per specified mission duration (5 years). For the electrons, the latter are also given in terms of

inner and outer zone contributions. Functionally derived differential fluxes are listed in the last columns for both species of particles.

Total orbit-integrated spectra in percent, for energy intervals  $\Delta E$  corresponding to the energy levels of the L-band tables, are also given in terms of average instantaneous and daily intensities.

An exposure index (for the normalization energies used in the L-band tables) is listed for nine successive intensity ranges varying by one order of magnitude, in terms of processed exposure duration (in hours) and total number of particles accumulated while in that intensity range for the indicated number of hours.

### III. Peaks and Totals Per Orbit: Tables 25-30

These tables contain the absolute instantaneous peak fluxes and the total fluences accumulated during each successive revolution, as obtained from the nominal trajectories for the investigated flight duration (24 hours of mission time).

Specifically, there are nine columns on these tables. Column 1 is an orbit counting device, based on:

- a) the orbit period when the trajectory is circular and lies in the equatorial plane;
- b) the physical perigee in all elliptical flight-path cases; and,
- c) the equatorial crossing for circular inclined trajectories.

Column 2 gives the peak flux. Columns 3, 4 and 5 indicate the spacecraft position in geocentric coordinates at which the predicted peak flux was encountered. Columns 6, 7 and 8 determine respectively the relative orbit time and the magnetic B-L coordinates for this event. For the purpose of orbital radiation studies, all simulated trajectories start at  $t_0 = 0$  hours. Finally, the last column indicates the total predicted flux to be encountered during that particular orbit. It is advisable to disregard the last line on this table because many times that orbit is incomplete and the fluxes or positions shown do not correspond to true peaks.

### IV. Time-Accounting and Exposure-Analysis: Tables 31-36

The "EXPOSURE-ANALYSIS" summary indicates what percent of its total lifetime T the satellite spends in "flux-free" regions of space, what percent of its total lifetime it spends in high intensity proton and electron domains, and while so exposed, what percent of its total flux it accumulates.

In the context of this study, the term "flux-free" applies to all regions of space where trapped particle fluxes are less than one proton or electron per square centimeter per second, having energies  $E > 5$  MeV, and  $E > .5$  MeV, respectively. By definition, this includes all regions external to the Van Allen radiation belts.

The concept of "trapped particle fluxes" is meant to include stably trapped, pseudo trapped, and transient fluxes, as long as they are part of or contained in the environment models used and, in the case of transients or pseudos, their sources are considered powerful enough to supply them continuously in substantial numbers.

Similarly, as "high intensity" are defined those regions of space where the instantaneous, integral, omnidirectional, trapped-particle flux is greater than  $10^3$  protons with energies  $E > 5$  MeV, and greater than  $10^5$  electrons with energies  $E > 5$  MeV.

The values given in these tables are statistical averages, obtained over extended intervals of mission time. However, they may vary significantly from one orbit to the next, when individual revolutions are considered.

The "TIME-ACCOUNT" breakdown shows what percent of its total time the satellite spends in the "inner zone" ( $1.0 \leq L < 2.8$ ) and in the "outer zone" ( $2.8 \leq L < 11.0$ ) electron trapping domains, and also the percent of time spent in regions external to the latter ( $L > 11.0$ ).

It should be noted that the confinement of the outer zone within the boundary of the  $L=11.0$  earth radii volume is arbitrary and has no physical meaning. It is intended only as a simplification to facilitate the calculations. The region considered "external" in this study ( $L > 11.0$ ) is still partially a domain of the outer zone, at least as far out as  $L=12.0$  earth radii, according to the current environment models.

A last item on this table: the inner zone time is further subdivided into two parts: the percentage of time spent outside ( $L < 1.1$ ) and inside ( $1.1 \leq L < 2.8$ ) the trapping domain.

#### V. Solar Proton Fluences and Exposure Factor: Tables 37-42

For the specified mission duration  $\tau$  (printed in the sub-title), and dipole cut-off shell ( $L=5$  earth radii, shown in the header), this table lists the solar proton fluence-spectra (in units of particles per square centimeter) at five discrete confidence levels  $Q$  (given at the top of each column).

The exposure factors (in percent of total mission duration) obtained from the geomagnetic shielding analysis are also listed for four dipole cut-off shell values (in earth radii).

#### VI. Total Dose and Components: Tables 43-60

These tables list doses in units of  $\text{rads}_{AL}$  as a function of aluminum shield thickness, given in three ways: range  $s$  in grams per square centimeter, depth  $t$  in millimeters, and depth  $t$  in mils.

Electron, bremsstrahlung, and proton contributions to the overall sumtotal dose are given separately. Electron and proton doses are further broken down into their respective constituents; namely, inner zone and outer zone for the former, trapped and solar flare for the latter.

The specific mission duration for which the doses have been calculated is indicated in the table headline.

Caution: the AL-event solar flare protons are not contributed gradually over the investigated mission duration ( $\tau = 5$  years) but are imparted in toto in a relatively short burst, that is, within approximately 2-4 days per AL event.

## B. Graphical Presentation

Some of the tabulated data are also plotted in Figures 1 to 36, and 49 to 102 with additional Figures 37 to 48 containing plots of flight path data. Positional flux and dose data are plotted in Figures 103-162. As with the tables, the computer plots are arranged in eight sets, where again each set pertains to one specific type of data. The first three sets have two similar members for each trajectory investigated: one for each particle species. The next two sets (fourth and fifth) contain one member for each trajectory considered. The sixth set contains nine similar members for each trajectory, providing three graphs (for respective depth ranges) for each of three geometries. The seventh set contains two similar members for each trajectory: one for each particle species. Finally, the last set has eight members for each trajectory.

### I. Time and Flux Histograms: Figures 1-12

These plots show two curves superimposed on the same graph; namely, one each for the variables "time" and "flux". Both are given on a semi-log scale as functions of the parameter  $L$  (earth radii), within the range  $1 < L \leq 10$ , and for constant  $L$ -bands of .1 earth radius width. The plots are plotted:

- a) by a plain curve, the characteristic trajectory intensities as obtained from the orbital integration process in terms of averaged integral particle fluxes above a given energy.
- b) by a contour marked with symbols, the percent total lifetime (%T) spent in each  $L$  interval.

The logarithmic ordinate relates to the time-flux variables. The printed numbers are powers of 10 and pertain to the fluxes; the scale values for the time curves are given in the upper part of the ordinate label: from  $10^{-3}$  to  $10^2$  percent of  $T$ , the type of particles, their integral energy, and the units, are all given in the lower part of the label. The label on the top of the graph identifies the trajectory.

### II. Spectral Profiles: Figures 13-24

A graphical presentation of the final composite spectral distribution, obtained from the orbital integration process. The plots are semi-log graphs, where the abscissa is a linear energy scale for integral particle energies  $E$ , in MeV, and the ordinate is a logarithmic scale for the fluxes, given in daily averages for energies greater than  $E$ ; the printed scale values are powers of 10.

### III. Peaks Per Orbit: Figures 25-36

Here the absolute peak intensities, encountered per period (1 period = 1 revolution = 1 orbit), are plotted for the duration of the flight-time processed in the analysis. The logarithmic ordinate, with scale values in powers of 10, relates to instantaneous particle fluxes of the environment at the indicated energy thresholds, while the abscissa is a linear orbit enumeration.

#### IV. Trajectory World Map Projections: Figures 37-42

These graphs depict the surface trace of the geocentrically projected subsatellite positions. The trajectories are plotted for several revolutions on a global map produced by a Miller Cylindrical Projection method. The contours of the continents have been omitted for clarity. The positions of equatorial crossing, of physical perigee, or of period commencement are indicated by numbers identifying the orbits shown in the graphs. For all trajectories, the distance between successive sequential numbers is a measure of the orbit precession.

#### V. Flight Path Tracing in B-L Space: Figures 43-48

Plots showing trajectory traces in B-L space on a semi-log scale. Several orbits are depicted, each identified by its sequential number. The magnetic equator is entered on all plots. The logarithmic ordinate relates to the field strength B in gauss; the printed values are exponents of 10. L is given in earth radii on the linear abscissa.

#### VI. Dose-Depth Curves by Geometry: Figures 49-102

Plots of final depth-dose values for the indicated mission duration. Normally, these plots show composite curves for bremsstrahlung, combined electrons (inner and outer zone), combined protons (trapped and solar flare), and sumtotal of all contributions. In the present case, the respective contours consist of inner and outer zone electrons separately and of trapped and solar flare protons, separately, of composite bremsstrahlung, and the sumtotal.

For ease of use and in order to provide a greater resolution at the more sensitive range of depths (namely the thinner shields) three plots have been generated per processed trajectory, for shield-ranges and subdivisions increasing by one order of magnitude.

The logarithmic ordinate, with scale values in powers of 10, relates to aluminum dose in units of rads. The linear abscissa is the shield thickness, given in three different units: range s in grams per square centimeter, depth t in millimeters, depth t in mils.

#### VII. Positional Flux Plots: Figures 103-114

Plots of instantaneous omnidirectional trapped particle fluxes (electrons and protons) at (up to 10) specified threshold energy levels (>MeV), for a selected orbit (usually worst case revolution through heart of SAA).

The logarithmic ordinate, with scale values in powers of 10, relates to the number of particles per square centimeter per second. The linear abscissa is the relative time, in minutes or hours, from the beginning of the selected orbital pass.

## VIII. Positional Dose Plots: Figures 115-162

Plots of instantaneous omnidirectional trapped particle dose values at (up to 10) specified shield thicknesses (omnidirectional isotropic incidence, cosine-theta distribution) for a selected orbit (usually worst case revolution through heart of SAA). Separate plots are generated (if present) for: electron dose (including bremsstrahlung), proton dose, and total dose (no solar proton contributions are included) for dose at transmission surface of aluminum slab shields, dose in semi-infinite aluminum medium, and dose at center of aluminum spheres.

The logarithmic ordinate, with scale values in powers of 10, relates to the respective dose in units of rads-aluminum. The linear abscissa is the relative time, in minutes or hours, from the beginning of the selected orbital pass.

### 11. RESULTS: ANALYSIS AND DISCUSSION

In this section, some of the presented tabular or graphical study-results are discussed, with occasional comments as to their use, limits, and applications.

#### A. Spectral Profiles

Characteristic features of the near earth radiation environment are strong altitude and inclination dependencies. However, at high inclination values ( $30^\circ < i < 90^\circ$ ) small changes in inclination will not produce changes in flux levels and spectral distributions as significant as those produced by small changes in altitude. The greatest inclination dependent variations occur in the range  $0^\circ \leq i \leq 30^\circ$ .

I. Protons: The protons exhibit relatively hard, almost uniform spectra in the investigated inclination-altitude regime.

It should be noted that a characteristic softening of the high energy tail of the spectrum appears at the 6389 km altitude orbit, near  $E > 400$  MeV, as a consequence of the limit of the volume of space occupied by these particles. The extent of the proton trapping domain along the magnetic equator is inversely related to their energy. This is strikingly demonstrated by the spectrum of the 10371 km altitude orbit, which falls off sharply at about  $E > 100$  MeV and which indicates that according to the standard model there are no protons with energies  $E > 150$  MeV contained in the region of space sampled by this trajectory.

II. Electrons: The electrons show complex variable spectra. Inner zone and outer zone average, orbit-integrated, composite intensities rise non-uniformly with altitude, particularly at energies above 3.75 MeV with differences reaching up to several orders of magnitude at  $E > 6$  MeV. Spectra extend to higher energies as height increases.

The inner zone spectra fall rapidly off to zero flux in the energy range from 4 to 5 MeV and they are therefore more benign than their harder outer-zone counterparts, which extend to energies of about 7 MeV.

At low altitudes (1667 and 2593 km), the inner zone contributions prevail up to about  $E > 2.75$  MeV, but for all energies  $E > 2$  MeV the inner zone spectra are always softer than the outer zone spectra (in agreement with the models).

However, at high altitudes (e.g. the 10371 km orbit), the inner zone contributions become insignificant even at low energies ( $.1 < E(\text{MeV}) < .7$ ).

#### B. Peaks Per Orbit

The absolute peaks per revolution have been obtained for standard processing energies:  $E > 5$  MeV for protons and  $E > .5$  MeV for electrons. Other energy selections produce different peak curves in an inverse relationship: lower energies yield higher and more expanded contours, and vice versa.

Peak contours of inclined circular trajectories display amplitude variations and sometimes discontinuities (flux-free time) that follow periodic patterns based on the daily cycle of revolutions. For fixed energies, amplitudes and discontinuities are function of: (a) inclination  $i$ , and (b) altitude  $h$ .

Variations in either  $i$  or  $h$  may produce significant changes in the amplitude of the peak curves and in the duration of the discontinuities: up to several orders of magnitude for the former, and completely eliminating the latter.

For the investigated trajectories at the given, fixed, inclination ( $i = 60^\circ$ ), the following observations can be made:

a) protons: at the processed energy of  $E > 5$  MeV only small variations are obtained for the low altitude orbits at 1667 and 2593 km; the difference between maximum and minimum peak intensity predicted is about a factor of 6 and 3, respectively. These differences disappear at higher altitudes, while at the same time the peak flux levels rise by about two orders of magnitude, from  $\sim 7 \times 10^4$  particles per square centimeter at 1667 km to  $\sim 7 \times 10^6$  particles per square centimeter at 5186 km.

b) electrons: at the processed energy of  $E > .5$  MeV the electron peaks appear very similar to those of the protons, as described in the previous paragraph. But the electron peak fluxes start at higher levels and reach an upper limit at the same altitude as the protons only with smaller differences: from  $\sim 3 \times 10^6$  particles per square centimeter at 1667 km to  $\sim 1 \times 10^7$  particles per square centimeter at 5186 km.

#### C. Flux-Free Time

Some comments on this topic have been provided in the previous section and in section 10/IV. Here a more detailed discussion will be given.

Flux-free time (FFT) intervals are an important feature of certain orbital configurations. They may occur over short orbit segments (partial FFT per period) or over the entire length of a revolution (total FFT per period). In terms of geomagnetic geometry, the FFTs establish the duration for which the trajectory lies outside the trapping domain of the corresponding particle species, evaluated at the given energies. Or conversely, they are a measure of the degree to which the trajectory is exposed to the charged particle trapping domains.



The number of consecutive flux-free orbits of circular trajectories is primarily a function of altitude and inclination and to a lesser degree a function of particle energy. Of the investigated flight paths, and for the selected energies (electrons:  $E > .5$  MeV, protons:  $E > 5$  MeV), none shows any completely flux-free revolutions per day. The total FFT, which derives completely from partially exposed revolutions (see "Exposure Analysis," Tables #31 to 36), in percent of total mission duration, can be summarized as follows:

<u>Altitude</u> (km)	<u>Protons</u> ( $E > 5$ MeV)	<u>Electrons</u> ( $E > .5$ MeV)
1667	22%	2%
2593	24%	3%
3889	30%	4%
5186	35%	6%
6389	39%	7%
10371	51%	11%

Higher energies will yield longer FFTs because the more energetic particles occupy a smaller volume of space.

#### D. Dose and Shielding

The calculated doses display features characteristic of the terrestrial radiation environment: at medium-to-high shield thicknesses, small contributions from relatively benign and low intensity electron spectra combined with major contributions from comparatively hard and intense proton spectra; at thin shield thicknesses the electrons predominate.

Depending on altitude, the proton doses prevail for shield thicknesses from greater than 16 to greater than 200 mils of aluminum. At high altitudes (e.g. 10371 km orbit) the opposite is true: protons are the major dose contributors only for shield thicknesses of less than 5 mils of aluminum.

Significant, however, is the fact that for aluminum the proton dose is only a weak function of shield thickness, as it shows very little attenuation over the evaluated depth range. Thus, in order to get an appreciable reduction in the dose, say by one order of magnitude, a 20-fold increase in shield thickness is necessary. The same is true for the bremsstrahlung dose. However, at low altitudes, in comparison to the proton contributions, the bremsstrahlung dose is so small (about 2 orders of magnitude lower) that it may be disregarded.

#### I. Decay and Degradation

The total doses obtained for each of the six investigated trajectories for the five year mission duration are substantial. In terms of electronics decay or materials degradation, the doses to be experienced on this mission inside the satellite, that is, behind a skin of about 80 mils of equivalent aluminum, are severe even for insensitive components or equipment:

<u>Altitudes (km)</u>	<u>Spherical Dose</u>
1667	~294K rads
2593	~1093K rads
3889	~1558K rads
5186	~1266K rads
6389	~1099K rads
10371	~1682K rads

## II. Contamination and Interference

It should be remembered, that the direct or indirect effects of the radiation environment may also be a nuisance in terms of instrument interference or measurement contamination. If such is the case, some remedies may be available.

## III. Possible Improvements

In the event that the magnitude of total dose or degree of radiation penetration behind the skin of the satellite is of importance to the mission, four possibilities exist to reduce the radiation effects on instrument and components:

- a) build or design an instrument less sensitive to radiation and construct the on-board and/or on-ground data processing software to remove or suppress radiation-induced noise
  - b) change the orbit by any combination of the elements eccentricity, altitude, and inclination so as to achieve a more benign environment
  - c) change the mission epoch: solar max for reduced proton intensities, solar min for reduced electron intensities
  - d) provide increased shielding either by geometry or by weight or by a combination of both;
    - by geometry: perform a 3-D analysis (solid angle sectoring) and rearrange other equipment on board the satellite in order to provide maximum protection to sensitive part over greatest possible fraction of solid angle.\*
    - by weight: place additional shields around sensitive part as needed.
- Clearly options (a), (b), and (d) are most readily accesible.

\*A powerful computer package for complex radiation shielding and transport calculations is now operational at GSFC. It is capable of addressing such topics as: (a) material mixtures, cross sections for protons, electrons, heavy charge particles, and neutrons, including source spectra and response functions; (b) source geometry, detector geometry, surfaces, rays, bodies, regions, body intersections, body unions, simple meshes, design bodies, spacecraft rays, with diverse features such as combinatorial options, translate-rotate-replicate capabilities, etc.; (c) heavy charged particle applications-1D transport by numerical integration, small volume pulse height (soft errors), 3D ray trace sectoring, 3D adjoint Monte Carlo; (d) electron bremsstrahlung-1D transport by numerical integration and by adjoint Monte

Carlo, small volume pulse height, 3D ray trace sectoring, 3D forward and adjoint Monte Carlo, energy deposition, charging distributions.

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ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VARIOUS AERIAL, AE6, AE17 FOR SOLAR MAXIMUM ELECTRONS (AE6) OF 1979  
 UNCERTAINTY FACTORS APPLIED FOR THIS RUN ARE: FOR PROTONS (AP6) - JF= 1.0; FOR INNER ZONE ELECTRONS (AE6) - UF= 1.0  
 MAGNETIC COORDINATES H AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG. MODEL 4; BARAKLUGH ET AL. 16-6-TRM 1975 \* TIME= 1989.5  
 MEMBLE. NXXLS-MA \* LATION CODES-DELETES-IGRAM AS ABOVE; 1472M AS BAUNALI TADE; IDSAAD, MA DELUDE 1.005  
 FOR INFORMATION OR EXPLANATION CONTACT: STASSINCPOLCS AT NASA-JSC, CUPE CVI, GREENBELT, MARYLAND 20771 JFL (301)-344-8067 \*  
 \*\*\*\*\*  
 SPECTRAL DISTRIBUTION: NORMALIZED BY FLUX OF ENERGY GREATER THAN 5.000MEV \*  
 \*\*\*\*\*  
 BANDA S

ENERGY LEVELS > (MEV)	1.0-1.2*	1.2-1.4*	1.4-1.6*	1.6-1.9*	1.9-2.3*	2.3-2.4*	2.4-2.8*	2.8-3.0*	3.0-3.2*	3.2-3.4*
1000	1.05E 00	1.16E 00	1.32E 00	1.75E 00	2.15E 00	2.77E 01	1.39E 02	6.97E 02	3.09E 03	1.67E 04
2000	1.02E 00	1.05E 00	1.19E 00	1.46E 00	2.15E 00	1.61E 00	7.51E 00	1.28E 01	1.51E 01	3.17E 01
5000	5.80E-01	9.03E-01	2.27E-01	6.54E-01	5.31E-01	3.62E-01	1.07E 00	1.07E 00	1.00E 00	1.00E 00
10000	6.45E-01	8.38E-01	7.24E-01	4.87E-01	1.21E-01	1.49E-01	2.57E-01	1.67E-01	1.44E-01	7.97E-02
25000	8.72E-01	5.69E-01	4.31E-01	1.82E-01	8.00E-02	1.69E-02	5.68E-03	2.42E-03	0.0	0.0
50000	7.46E-01	4.20E-01	3.02E-01	1.06E-01	3.01E-02	7.20E-03	2.14E-03	5.81E-04	0.0	0.0
100000	9.43E-01	7.59E-01	1.77E-01	4.38E-02	7.63E-03	1.39E-03	0.0	0.0	0.0	0.0
500000	2.32E-02	6.25E-03	2.23E-03	8.98E-05	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX#	1.06E 06	2.64E 08	1.57E 08	6.44E 07	3.40E 07	2.17E 07	5.46E 06	1.51E 06	4.14E 05	1.25E 05

ENERGY LEVELS > (MEV)	3.4-3.6*	3.6-3.8*	3.8-4.0*	4.0-4.2*	4.2-4.4*	4.4-4.6*	4.6-4.8*	4.8-5.3*	5.3-5.4*	5.4-5.6*	5.6-5.8*
1000	3.85E 05	1.98E 07	2.10E 07	1.13E 07	6.54E 06	2.23E 06	2.44E 06	4.41E 06	2.41E 06	3.69E 06	1.06E 06
2000	1.78E 02	3.22E 03	2.62E 03	6.99E 02	2.39E 02	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX#	6.45E 03	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS > (MEV)	5.8-6.0*	6.0-6.2*	6.2-6.4*	6.4-6.6*	6.6-6.8*	6.8-7.0*	7.0-7.2*	7.2-7.4*	7.4-7.6*	7.6-7.8*	7.8-8.3*	8.3-8.2*
1000	3.25E 05	3.20E 05	2.25E 04	1.49E 04	3.09E 04	3.62E 03	1.25E 02	3.65E 00	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX#	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS > (MEV)	8.2-8.4*	8.4-8.6*	8.6-8.8*	8.8-9.0*	9.0-9.2*	9.2-9.4*	9.4-9.6*	9.6-10.0*	10.0-10.2*	10.2-10.4*	10.4-10.8*	10.8-11.4*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX#	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

TABLE 1









\*\*\*\*\*  
 ORBITAL FLUX STUDY WITH COMPENSATE PARTICLE ENVIRONMENTS: VETTES ADP; A26, A517 FOR SOLAR MAXIMUM  
 \*\* DIRECTIONALITY FACTORS (DF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APR 8), DF= 1.0; FOR IONER ZONE FLECTIONS (APR 8) DF= 1.0; UNIFLUX OF 1979  
 \*\* MAGNETIC COORDINATES (B) AND L COMPUTED BY IYAS EA OF 1972 WITH ALL MAG A MODEL 3833K \*\* B/L FROM 188-109 TO 168-109 \*\* TIME PERIOD 1989.5  
 \*\* VEHICLE : NVL315-MAX \*\* INCLINATION = 0 DEG \*\* PERIGEE = 389 KM \*\* MODEL 3833K \*\* B/L FROM 188-109 TO 168-109 \*\* TIME PERIOD 1989.5  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT S.G. STASSINOPOULOS AT NASA-GSFC, CODE 011, GREENBELT, MARYLAND 20771, TEL. (301)-344-8067  
 \*\*\*\*\*

ENERGY LEVELS > (MEV)	L - B A M D S *	M A G N E T I C *	S H E L L *	P A R A M E T E R *	I N F A R I *	F A R I *	P A R I *	L - B A M D S
1000	0.0	0.0	2.20E 00	1.31E 01	5.20E 01	1.90E 02	9.94E 02	1.53E 04
2000	0.0	1.32E 00	1.88E 00	3.74E 00	1.71E 00	1.40E 01	1.43E 01	1.68E 03
3000	0.0	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00	1.00E 00
4000	0.0	6.57E-01	4.36E-01	3.43E-01	2.35E-01	1.77E-01	1.30E-01	9.21E-02
5000	0.0	4.85E-02	3.21E-02	2.50E-02	1.83E-02	1.40E-02	1.05E-02	7.81E-03
1000.0	0.0	1.39E-02	1.59E-02	1.82E-02	2.50E-02	3.07E-02	4.24E-02	6.00E-02
500.0	0.0	1.92E-04	5.20E-05	3.48E-05	2.31E-05	1.52E-05	9.82E-06	6.52E-06
MORFLUX=	0.0	1.99E 09	9.45E 09	1.82E 09	5.24E 08	6.37E 07	2.04E 07	6.07E 06
ENERGY LEVELS > (MEV)	L - B A M D S * <th>M A G N E T I C *</th> <th>S H E L L *</th> <th>P A R A M E T E R *</th> <th>I N F A R I *</th> <th>F A R I *</th> <th>P A R I *</th> <th>L - B A M D S</th>	M A G N E T I C *	S H E L L *	P A R A M E T E R *	I N F A R I *	F A R I *	P A R I *	L - B A M D S
1000	2.44E 05	1.18E 06	2.68E 08	1.10E 08	7.97E 07	5.40E 07	4.30E 07	1.06E 07
2000	3.07E 00	2.99E 04	0.0	0.0	3.35E 02	0.0	0.0	0.0
3000	1.00E 00	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	4.61E-03	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1000.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
MORFLUX=	1.42E 05	1.67E 04	0.0	0.0	0.0	0.0	0.0	0.0
ENERGY LEVELS > (MEV)	L - B A M D S * <th>M A G N E T I C *</th> <th>S H E L L *</th> <th>P A R A M E T E R *</th> <th>I N F A R I *</th> <th>F A R I *</th> <th>P A R I *</th> <th>L - B A M D S</th>	M A G N E T I C *	S H E L L *	P A R A M E T E R *	I N F A R I *	F A R I *	P A R I *	L - B A M D S
1000	6.85E 06	4.30E 06	5.06E 06	1.12E 06	6.72E 05	5.58E 04	6.70E 03	1.63E 03
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1000.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
MORFLUX=	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
ENERGY LEVELS > (MEV)	L - B A M D S * <th>M A G N E T I C *</th> <th>S H E L L *</th> <th>P A R A M E T E R *</th> <th>I N F A R I *</th> <th>F A R I *</th> <th>P A R I *</th> <th>L - B A M D S</th>	M A G N E T I C *	S H E L L *	P A R A M E T E R *	I N F A R I *	F A R I *	P A R I *	L - B A M D S
1000	1.54E 00	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1000.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
MORFLUX=	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

TABLE 5

ORBITAL PLOX STUDY WITH COMPOSITE PARTICLE ENVIRONMENT: VETTES APB AEB, AEF FOR SCAB MAXIMUM UNIFLUX OF 1977  
 UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN AREA: FOR POTIONS (APB) - UF= 1.0 FOR INHER ZONE ELECT IONS (AEB) - UF= 1.0  
 MAGNETIC COORDINATES B AND L COMPUTED BY INVADA 1972 \*INVAHALL MAG NODE= 4883KM \*B L CABBIT TAPE: I03446 \*PLOT CODE= 2876  
 VERBIC: MAX \*INCLINATION= 60DEG \*PRIGGE= 389KM \*APOSE= 3883KM \*D L CABBIT TAPE: I03446 \*PLOT CODE= 2876  
 FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINPOULOS AT NASA-USFCC CODE G011 GREENBELT, MARYLAND 20771 TEL. (301)-344-8067

SPECTRAL DISTRIBUTION: NORMALIZED BY FLOW OF ENERGY GREATER THAN .500CFV

ENERGY LEVELS > (HEV)	1.0-1.2	1.2-1.4	1.4-1.6	1.6-1.8	1.8-2.0	2.0-2.2	2.2-2.4	2.4-2.6	2.6-2.8	2.8-3.0	3.0-3.2	3.2-3.4
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
MORFLUX=	0.0	2.472	1.072	1.072	2.162	4.272	1.052	4.572	1.832	2.922	7.822	1.352
ENERGY LEVELS > (HEV)	3.0-3.6	3.6-4.2	4.2-4.8	4.8-5.4	5.4-6.0	6.0-6.6	6.6-7.2	7.2-7.8	7.8-8.4	8.4-9.0	9.0-9.6	9.6-10.2
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
MORFLUX=	1.972	1.832	2.492	2.002	1.922	1.602	1.172	1.032	8.482	3.122	3.512	1.462
ENERGY LEVELS > (HEV)	10.0-12.0	12.0-14.0	14.0-16.0	16.0-18.0	18.0-20.0	20.0-22.0	22.0-24.0	24.0-26.0	26.0-28.0	28.0-30.0	30.0-32.0	32.0-34.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
MORFLUX=	1.822	1.312	1.882	5.812	5.122	4.322	2.412	1.982	1.482	1.192	2.002	3.052
ENERGY LEVELS > (HEV)	30.0-40.0	40.0-50.0	50.0-60.0	60.0-70.0	70.0-80.0	80.0-90.0	90.0-100.0	100.0-110.0	110.0-120.0	120.0-130.0	130.0-140.0	140.0-150.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
MORFLUX=	1.102	4.472	5.072	2.322	1.622	1.942	1.642	5.842	5.622	1.542	1.462	0.0

TABLE 6

\*\*\*\*\*  
 \*\* JONIAL FLUX STUDY WITH COMPOSITE PARTICLE ENRICHMENTS: VITIES APPO, ABO, A-17 FOR SOLA MAXIMUM UNITS OF 1979  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (AP) = 1.0, FOR ALPHA = 1.0, FOR NEUTRONS = 1.0  
 \*\* MAGNETIC COORDINATES H AND L CALCULATED BY INVADE OF 1972 WITH ALLMAG MAGNETIC FIELD DATA FROM 1975 TO 1979.5  
 \*\* GEOMAGNETIC INDEX K<sub>p</sub> MAX \*\* INCLINATION \*\* COEG \*\* RETRIEVE FROM APPOHTE ELECTRONIC TABLES: 1975-1979.5  
 \*\* FOR INFORMATION OF EXPLANATION OF CONTACT LOG, STAFF CONTACTS AT NASA-GSFC: G. J. COLEMAN, 27711 L. (301)-343-3067  
 \*\*\*\*\*  
 \*\* SPECTRAL DISTRIBUTION: NORMALIZED BY FLUX OF ENERGY AT 1000.000 V \*\*\*\*\*  
 \*\*\*\*\*

ENERGY LEVELS >(MEV)

ENERGY LEVELS >(MEV)	1.0-1.20	1.2-1.40	1.4-1.60	1.6-1.80	1.8-2.00	2.0-2.20	2.2-2.40	2.4-2.60	2.6-2.80	2.8-3.00	3.0-3.20	3.2-3.40
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS >(MEV)

ENERGY LEVELS >(MEV)	1.0-1.20	1.2-1.40	1.4-1.60	1.6-1.80	1.8-2.00	2.0-2.20	2.2-2.40	2.4-2.60	2.6-2.80	2.8-3.00	3.0-3.20	3.2-3.40
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS >(MEV)

ENERGY LEVELS >(MEV)	1.0-1.20	1.2-1.40	1.4-1.60	1.6-1.80	1.8-2.00	2.0-2.20	2.2-2.40	2.4-2.60	2.6-2.80	2.8-3.00	3.0-3.20	3.2-3.40
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS >(MEV)

ENERGY LEVELS >(MEV)	1.0-1.20	1.2-1.40	1.4-1.60	1.6-1.80	1.8-2.00	2.0-2.20	2.2-2.40	2.4-2.60	2.6-2.80	2.8-3.00	3.0-3.20	3.2-3.40
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMFLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

TABLE 7



\*\*\*\*\* UNIFLUX OF 1979 \*\*\*\*\*  
 \*\* JERITA FLUX STUDY WITH SAMPLE PARTICLE ENHANCEMENTS: VLTIES ARE: ALS, AL17 FOR SLM, MAXIMUM \*\*  
 \*\* UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FOR FRCTIONS (FRS) - U= 1.0; FOR INIT ZPRL ELECTRON (AE) - U= 1.0  
 \*\* MAGNETIC COORDINATES (U) AND COMPUTATION BY INAVPA OF 1972 WITH ALL MAG. MODEL 4; FOR ACADZ FOR ITAL 16-TPM 1975 - TIME 1989.45 \*\*  
 \*\* AT RIGLE 63424 \*\* AFFICE 6285KM \*\* ZL UNIT TAPE: 105442 \*\* PERICE 3.988 \*\*  
 \*\* FOR INFORMATION OF EXPLANATION CONTACT THE STATIONING OFFICE AT NADA-CR-COJL POLY CENTER, OAKLAND 20771, TEL: (510) 344-4067 \*\*  
 \*\* \*\* \*\* \*\*

\*\*\*\*\* SPECTRAL DISTRIBUTION: NORMALIZED BY FLUX OF ENERGY CARRIER THAT OCCURRES \*\*\*\*\*  
 \*\* \*\* \*\*

ENERGY LEVELS >(MEV)	1.0-1.2*	1.2-1.4*	1.4-1.6*	1.6-1.8*	1.8-2.0*	2.0-2.2*	2.2-2.4*	2.4-2.6*	2.6-2.8*	2.8-3.0*	3.0-3.2*	3.2-3.4*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMAL FLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS >(MEV)	1.0-1.2*	1.2-1.4*	1.4-1.6*	1.6-1.8*	1.8-2.0*	2.0-2.2*	2.2-2.4*	2.4-2.6*	2.6-2.8*	2.8-3.0*	3.0-3.2*	3.2-3.4*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMAL FLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS >(MEV)	1.0-1.2*	1.2-1.4*	1.4-1.6*	1.6-1.8*	1.8-2.0*	2.0-2.2*	2.2-2.4*	2.4-2.6*	2.6-2.8*	2.8-3.0*	3.0-3.2*	3.2-3.4*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMAL FLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

ENERGY LEVELS >(MEV)	1.0-1.2*	1.2-1.4*	1.4-1.6*	1.6-1.8*	1.8-2.0*	2.0-2.2*	2.2-2.4*	2.4-2.6*	2.6-2.8*	2.8-3.0*	3.0-3.2*	3.2-3.4*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
10000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
25000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
50000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
100000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
500000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NORMAL FLUX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

TABLE 9

\*\*\*\*\* ORBITAL FLUX STUDY, MILLENNIUMS, PARTIAL ENVIRONMENT: VETTES (SPH: AEP, AET) FOR 2001, MILLENNIUM \*\*\*\*\* SHELLX OF 1979 \*\*\*\*\*  
 \*\* IN CERTAIN ENERGY INTERVALS (J) APPLICABLE FOR VETTES (SPH: AEP, AET) FOR 2001, MILLENNIUM \*\*\*\*\* SHELLX OF 1979 \*\*\*\*\*  
 \*\* MAGNETIC (CORE) INAVALS (J) COMPUTED BY CURVA (C) OF 1972 WITH ALL VETTES (SPH: AEP, AET) FOR 2001, MILLENNIUM \*\*\*\*\* SHELLX OF 1979 \*\*\*\*\*  
 \*\* FOR MORE INFORMATION OR FOR AN ANNOTATION CONTACT THE STAFF OF THE STATION AT NASA-CR/C-CODE 311, UNIVERSITY OF MARYLAND, 20771 TEL: (301)-344-8067 \*\*\*\*\*  
 \*\* SPECIAL DISTRIBUTION: GENERALIZED BY FLUX OF ENERGY (MEV) FOR 2001, MILLENNIUM \*\*\*\*\* SHELLX OF 1979 \*\*\*\*\*

ENERGY LEVELS >(MEV)	1.0-1.2*	1.2-1.4*	1.4-1.6*	1.6-1.8*	1.8-2.0*	2.0-2.2*	2.2-2.4*	2.4-2.6*	2.6-2.8*	2.8-3.0*	3.0-3.2*	3.2-3.4*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1500	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NCFMFLUX= 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0												
ENERGY LEVELS >(MEV)	3.4-3.6*	3.6-3.8*	3.8-4.0*	4.0-4.2*	4.2-4.4*	4.4-4.6*	4.6-4.8*	4.8-5.0*	5.0-5.2*	5.2-5.4*	5.4-5.6*	5.6-5.8*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1500	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NCFMFLUX= 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0												
ENERGY LEVELS >(MEV)	5.8-6.0*	6.0-6.2*	6.2-6.4*	6.4-6.6*	6.6-6.8*	6.8-7.0*	7.0-7.2*	7.2-7.4*	7.4-7.6*	7.6-7.8*	7.8-8.0*	8.0-8.2*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1500	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NCFMFLUX= 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0												
ENERGY LEVELS >(MEV)	8.2-8.4*	8.4-8.6*	8.6-8.8*	8.8-9.0*	9.0-9.2*	9.2-9.4*	9.4-9.6*	9.6-9.8*	9.8-10.0*	10.0-10.2*	10.2-10.4*	10.4-10.6*
1000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1500	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6000	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
NCFMFLUX= 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0												

TABLE 10

GEBHA FLUX STUDY WITH SUBSILE PARTICLES ENHANCEMENTS. MATTES AND M. DEY FOR ST. LOUIS UNIVERSITY. SNIFLUX OF 1979  
 \*\* JNC-5 TITANIC FACTORY (30) APPLIED FOR THIS RUN APRIL 2, 1979. THIS RUN WAS FOR TESTING PURPOSES. LOCATION: GEBHA 1.0  
 \*\* MAGNETIC COORDINATES: NAD 1983, UTM ZONE 18, Easting 550000, Northing 4900000. DATE: 1979.04.02. TIME: 19:00.5  
 \*\* FUM (INTEGRATION) : EXPLANATION : STASIS (FLUXES AT NAD-CENTERS) FOR L1, L2, L3, L4, L5, L6, L7, L8, L9, L10, L11, L12, L13, L14, L15, L16, L17, L18, L19, L20, L21, L22, L23, L24, L25, L26, L27, L28, L29, L30, L31, L32, L33, L34, L35, L36, L37, L38, L39, L40, L41, L42, L43, L44, L45, L46, L47, L48, L49, L50, L51, L52, L53, L54, L55, L56, L57, L58, L59, L60, L61, L62, L63, L64, L65, L66, L67, L68, L69, L70, L71, L72, L73, L74, L75, L76, L77, L78, L79, L80, L81, L82, L83, L84, L85, L86, L87, L88, L89, L90, L91, L92, L93, L94, L95, L96, L97, L98, L99, L100.

ENERGY LEVELS  
 1000 3.0  
 2000 3.0  
 3000 3.0  
 4000 3.0  
 5000 3.0  
 6000 3.0  
 7000 3.0  
 8000 3.0  
 9000 3.0  
 10000 3.0

NEUTRONS  
 1.7E+01 2.7E+01 3.7E+01 4.7E+01 5.7E+01 6.7E+01 7.7E+01 8.7E+01 9.7E+01  
 1.0E+02 2.0E+02 3.0E+02 4.0E+02 5.0E+02 6.0E+02 7.0E+02 8.0E+02 9.0E+02  
 1.0E+03 2.0E+03 3.0E+03 4.0E+03 5.0E+03 6.0E+03 7.0E+03 8.0E+03 9.0E+03  
 1.0E+04 2.0E+04 3.0E+04 4.0E+04 5.0E+04 6.0E+04 7.0E+04 8.0E+04 9.0E+04  
 1.0E+05 2.0E+05 3.0E+05 4.0E+05 5.0E+05 6.0E+05 7.0E+05 8.0E+05 9.0E+05  
 1.0E+06 2.0E+06 3.0E+06 4.0E+06 5.0E+06 6.0E+06 7.0E+06 8.0E+06 9.0E+06  
 1.0E+07 2.0E+07 3.0E+07 4.0E+07 5.0E+07 6.0E+07 7.0E+07 8.0E+07 9.0E+07  
 1.0E+08 2.0E+08 3.0E+08 4.0E+08 5.0E+08 6.0E+08 7.0E+08 8.0E+08 9.0E+08  
 1.0E+09 2.0E+09 3.0E+09 4.0E+09 5.0E+09 6.0E+09 7.0E+09 8.0E+09 9.0E+09  
 1.0E+10 2.0E+10 3.0E+10 4.0E+10 5.0E+10 6.0E+10 7.0E+10 8.0E+10 9.0E+10

ENERGY LEVELS  
 1000 3.0  
 2000 3.0  
 3000 3.0  
 4000 3.0  
 5000 3.0  
 6000 3.0  
 7000 3.0  
 8000 3.0  
 9000 3.0  
 10000 3.0

NEUTRONS  
 1.7E+01 2.7E+01 3.7E+01 4.7E+01 5.7E+01 6.7E+01 7.7E+01 8.7E+01 9.7E+01  
 1.0E+02 2.0E+02 3.0E+02 4.0E+02 5.0E+02 6.0E+02 7.0E+02 8.0E+02 9.0E+02  
 1.0E+03 2.0E+03 3.0E+03 4.0E+03 5.0E+03 6.0E+03 7.0E+03 8.0E+03 9.0E+03  
 1.0E+04 2.0E+04 3.0E+04 4.0E+04 5.0E+04 6.0E+04 7.0E+04 8.0E+04 9.0E+04  
 1.0E+05 2.0E+05 3.0E+05 4.0E+05 5.0E+05 6.0E+05 7.0E+05 8.0E+05 9.0E+05  
 1.0E+06 2.0E+06 3.0E+06 4.0E+06 5.0E+06 6.0E+06 7.0E+06 8.0E+06 9.0E+06  
 1.0E+07 2.0E+07 3.0E+07 4.0E+07 5.0E+07 6.0E+07 7.0E+07 8.0E+07 9.0E+07  
 1.0E+08 2.0E+08 3.0E+08 4.0E+08 5.0E+08 6.0E+08 7.0E+08 8.0E+08 9.0E+08  
 1.0E+09 2.0E+09 3.0E+09 4.0E+09 5.0E+09 6.0E+09 7.0E+09 8.0E+09 9.0E+09  
 1.0E+10 2.0E+10 3.0E+10 4.0E+10 5.0E+10 6.0E+10 7.0E+10 8.0E+10 9.0E+10

ENERGY LEVELS  
 1000 3.0  
 2000 3.0  
 3000 3.0  
 4000 3.0  
 5000 3.0  
 6000 3.0  
 7000 3.0  
 8000 3.0  
 9000 3.0  
 10000 3.0

NEUTRONS  
 1.7E+01 2.7E+01 3.7E+01 4.7E+01 5.7E+01 6.7E+01 7.7E+01 8.7E+01 9.7E+01  
 1.0E+02 2.0E+02 3.0E+02 4.0E+02 5.0E+02 6.0E+02 7.0E+02 8.0E+02 9.0E+02  
 1.0E+03 2.0E+03 3.0E+03 4.0E+03 5.0E+03 6.0E+03 7.0E+03 8.0E+03 9.0E+03  
 1.0E+04 2.0E+04 3.0E+04 4.0E+04 5.0E+04 6.0E+04 7.0E+04 8.0E+04 9.0E+04  
 1.0E+05 2.0E+05 3.0E+05 4.0E+05 5.0E+05 6.0E+05 7.0E+05 8.0E+05 9.0E+05  
 1.0E+06 2.0E+06 3.0E+06 4.0E+06 5.0E+06 6.0E+06 7.0E+06 8.0E+06 9.0E+06  
 1.0E+07 2.0E+07 3.0E+07 4.0E+07 5.0E+07 6.0E+07 7.0E+07 8.0E+07 9.0E+07  
 1.0E+08 2.0E+08 3.0E+08 4.0E+08 5.0E+08 6.0E+08 7.0E+08 8.0E+08 9.0E+08  
 1.0E+09 2.0E+09 3.0E+09 4.0E+09 5.0E+09 6.0E+09 7.0E+09 8.0E+09 9.0E+09  
 1.0E+10 2.0E+10 3.0E+10 4.0E+10 5.0E+10 6.0E+10 7.0E+10 8.0E+10 9.0E+10

TABLE 11







\*\*\*\*\* ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VELLIS, APB1, ALG1, ALI7 FOR SOLAR MAXIMUM LINE LUX OF 1979 \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: JF= 1.0; FOR INNER ZONE ELECTRICNS (AEG) - UF= 1.0  
 \*\* MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALLMAG, MODEL 4; BARRACLOUGH ET AL. 164-TRM 1975 \* TIME= 1989.5  
 \*\* VEHICLE: NM-115-MAY \*\* INCLINATION: CODEC \*\* PERIGEE: 16622M \*\* APOGEE: 16622M \*\* ORBIT TARE: 305AAG \*\* PERIOD: 1.005 \*\*  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINPOULOS AT NASA-GSFC, CODE 601, GREENHILL, MARYLAND 20771, TEL. (301) 344-8067 \*\*  
 \*\*\*\*\* ELECTRICNS \*\*\*\*\*

\*\*\*\*\* SPECTRUM IN PERCENT DELTA ENERGY \*\*\*\*\* \*\* COMPOSITE ORBIT SPECTRUM \*\*\*\*\* TAU= 5.00000 YR(S)

ENERGY RANGES >(MEV)	AVERAGED TOTAL FLUX #/CM**2/SEC	AVERAGED SPECTRUM TOTAL FLUX #/CM**2/DAY	PERCENT	ENERGY LEVELS >(MEV)	AVERAGED INTLG. FLUX #/CM**2/SEC	INTEGR. FLUX #/CM**2/DAY	AVERAGED TOTAL #/CM**2/TAU	INNER ZONE + OUTER ZONE #/CM**2/TAU	INTEGRAL #/CM**2/TAU	AVERAGED FLUX #/CM**2/SEC	DIFFERENTIAL FLUX #/CM**2/SEC
1.000-5.000	1.038E 07	8.894E 11	47.026	4.000E 01	6.116E 07	1.332E 11	2.386E 15	2.314E 15	6.030E 13	6.030E 04	6.030E 04
5.000-1.000	2.698E 05	2.321E 10	1.783	7.000E 01	1.205E 07	1.093E 12	1.595E 15	1.540E 15	5.523E 13	7.710E 04	7.710E 04
1.000-1.500	3.735E 04	3.227E 09	0.247	1.000E 00	1.062E 07	9.172E 11	1.674E 15	1.630E 15	4.425E 13	6.937E 04	6.937E 04
1.500-2.000	1.367E 04	1.162E 09	0.089	2.000E 00	4.920E 06	4.251E 11	7.750E 14	7.490E 14	2.720E 13	3.909E 04	3.909E 04
2.000-3.000	8.336E 03	7.202E 08	0.355	3.000E 00	2.085E 06	1.802E 11	3.289E 14	3.121E 14	1.680E 13	1.795E 04	1.795E 04
3.000-4.000	1.901E 03	1.652E 08	0.013	4.000E 00	8.124E 05	7.119E 10	1.281E 14	1.150E 14	1.251E 13	6.832E 03	6.832E 03
4.000-5.000	3.229E 02	2.750E 07	0.002	5.000E 00	3.312E 05	2.862E 10	5.222E 13	4.293E 13	5.302E 12	2.470E 03	2.470E 03
5.000-6.000	1.873E 01	1.618E 06	0.300	6.000E 00	2.127E 05	1.838E 10	3.354E 13	2.560E 13	7.937E 12	1.029E 03	1.029E 03
6.000-OVER	1.461E 02	1.243E 03	0.300	7.000E 00	1.415E 05	1.223E 10	2.331E 13	1.553E 13	2.782E 12	5.254E 02	5.254E 02
TOTAL	1.062E 07	9.172E 11	70.104	8.000E 00	1.023E 05	8.839E 09	1.613E 13	1.013E 13	5.802E 12	3.135E 02	3.135E 02
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				9.000E 00	7.892E 04	6.818E 09	1.244E 13	7.473E 12	4.565E 12	2.045E 02	2.045E 02
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				1.000E 00	8.481E 04	3.308E 09	9.693E 12	5.420E 12	4.200E 12	1.401E 02	1.401E 02
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				1.500E 00	3.832E 04	3.311E 09	5.042E 12	3.250E 12	2.784E 12	7.291E 01	7.291E 01
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				1.500E 00	2.835E 04	2.078E 09	3.793E 12	1.964E 12	1.829E 12	4.251E 01	4.251E 01
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				1.750E 00	1.593E 04	1.376E 09	2.512E 12	1.303E 12	1.208E 12	2.585E 01	2.585E 01
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				2.000E 00	1.959E 04	1.402E 08	1.668E 12	8.653E 11	6.025E 11	1.596E 01	1.596E 01
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				2.250E 00	7.624E 03	6.582E 08	1.202E 12	5.274E 11	6.148E 11	1.109E 01	1.109E 01
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				2.500E 00	5.528E 03	4.776E 08	8.716E 11	3.989E 11	4.727E 11	4.443E 00	4.443E 00
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				3.000E 00	3.378E 03	2.919E 08	5.226E 11	1.680E 11	3.644E 11	6.128E 00	6.128E 00
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				3.500E 00	2.442E 03	1.937E 08	3.439E 11	7.180E 10	2.817E 11	4.129E 00	4.129E 00
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				3.500E 00	1.313E 03	1.135E 08	2.071E 11	2.161E 10	1.859E 11	2.567E 00	2.567E 00
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				3.500E 00	5.155E 02	7.057E 07	1.288E 11	6.513E 09	1.223E 11	1.532E 00	1.532E 00
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				4.000E 00	5.241E 02	4.524E 07	3.264E 11	1.966E 09	2.368E 11	1.309E 00	1.309E 00
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				4.500E 00	3.416E 02	2.951E 07	5.348E 10	5.863E 08	5.324E 10	7.311E 01	7.311E 01
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				4.500E 00	7.572E 01	6.880E 06	1.257E 10	3.031E 07	1.254E 10	2.480E 01	2.480E 01
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				5.000E 00	1.875E 01	1.623E 06	2.556E 09	0.0	2.556E 09	7.008E 02	7.008E 02
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				5.000E 00	1.898E 00	1.640E 05	2.593E 08	0.0	2.593E 08	6.878E 03	6.878E 03
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				6.000E 00	1.401E 02	1.263E 05	2.304E 08	0.0	2.304E 08	1.324E 04	1.324E 04
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				6.500E 00	0.0	0.0	0.0	0.0	0.0	0.0	0.0
***** EXPOSURE INDEX: ENERGY>5.000 MEV *****				7.000E 00	0.0	0.0	0.0	0.0	0.0	0.0	0.0
TOTAL	24.000	2.462E 10		7.000E 00	0.0	0.0	0.0	0.0	0.0	0.0	0.0

TABLE 14

\*\*\*\*\*  
 \*\* ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTAL VELLS. APT: AELT FOR SOLAR MAXIMUM \*\*\*\*\* UNIFLUX OF 1979 \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FOR PROTONS (AP7) - UF = 1.3; FOR INNER ZONE ELECTRICONS (AL6) - UF = 1.3 \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES R AND L COMPUTED BY INVARA OF 1972 WITH ALLMAO, MODEL 4: BARRACLOUGH CTAL, ICR-TM 1975 # TIME = 1989.5 \*\*\*\*\*  
 \*\* VEHICLE - NALX22-SMAX. AN INCLINATION CODEC. RA BEGINS - 59.3 KM. RA ADGES - 25.3 KM. RA HLT LABEL: 305A6. RA BEGINS 2.339 \*\*\*\*\*  
 \*\* FOR INFORMATION OF EXPLANATION CONTACT E.C. STASSINPOULS AT NASA-GFC, CODE (31) GFC/PT/ST, MAIL YLANS 20771-1FL (3311)-348-8067 \*\*\*\*\*  
 \*\*\*\*\* PROTONS \*\*\*\*\*  
 \*\*\*\*\*

\*\*\*\*\* SPECTRUM IN PERCENT DELTA ENERGY \*\*\*\*\* \*\* COMPOSITE CRBIT SPECTRUM \*\* TAUF 5.0000 YEAR(S) \*\*\*\*\*

ENERGY RANGES >(MEV)	AVERAGED TOTAL FLUX #/CM**2/SEC	AVERAGED TOTAL FLUX #/CM**2/DAY PERCENT	ENERGY LEVELS >(MEV)	AVERAGED INTENSIFLUX #/CM**2/SEC	AVERAGED INTENSIFLUX #/CM**2/TAU	AVERAGED INTENSIFLUX #/CM**2/SEC/KEV
1.000-2.000	5.24E 05	7.52E 10	61.418	4.03E-011.504L 06	1.330E 11	2.372E 14
2.000-3.000	1.31E 04	1.14E 05	0.860	1.00E 001.897L 05	1.54E 10	1.37E 13
3.000-4.000	6.34E 03	5.54E 04	0.428	5.00E 007.652L 05	1.73E 10	6.26E 02
4.000-5.000	3.04E 03	2.75E 04	0.245	2.00E 004.501L 04	1.03E 10	1.30E 02
5.000-10.000	3.04E 03	2.75E 04	0.245	3.00E 005.441L 04	5.71E 09	2.94E 01
10.000-20.000	2.09E 03	1.81E 04	0.142	2.00E 004.703L 04	8.58E 08	9.62E 00
20.000-30.000	2.09E 03	1.81E 04	0.142	3.00E 004.703L 04	7.41E 08	6.85E 00
30.000-40.000	3.98E 01	3.44E 02	0.003	4.00E 004.144L 04	6.51E 08	5.14E 00
40.000-50.000	3.98E 01	3.44E 02	0.003	5.00E 004.680L 04	3.10E 08	2.45E 00
50.000-OVER	3.98E 01	3.44E 02	0.003	6.00E 004.680L 04	5.81E 08	4.65E 00
TOTAL	5.819E 05	8.544E 10	65.739	2.447E 09	4.447E 12	3.540E 00
				2.177E 08	1.481E 09	2.424E 00
				1.430E 08	1.213E 09	1.127E 00
				9.420E 03	8.024E 08	5.740E -01
				7.830E 01	6.740E 08	3.063E -01
				6.534E 03	5.643E 08	1.767E -01
				5.023E 03	5.203E 08	1.211E -01
				3.512E 03	4.262E 08	5.224E -02
				5.047E 03	4.361E 08	8.50E -02
				4.624E 03	3.995E 08	7.02E -02
				3.093E 03	3.536E 08	5.02E -02
				3.214E 03	2.775E 08	3.840E -02
				2.525E 03	2.182E 08	2.982E -02
				1.424E 03	1.233E 08	1.60E -02
				3.047E 02	6.970E 07	8.78E -03
				4.814E 02	4.236E 07	5.00E -03
				2.543E 02	2.542E 07	2.56E -03
				1.781E 02	1.579E 07	1.78E -03
				1.066E 02	9.331E 06	1.06E -03
				3.981E 01	3.443E 06	3.571E -04
TOTAL	2.4000	3.580E 09				

\*\*\*\*\* EXPOSURE INDEX: ENERGY>5.000 MEV \*\*\*\*\*

INTENSITY RANGES #/CM\*\*2/SEC (HOURS)

EXPOSURE DURATION (HOURS)	EXPOSURE TOTAL # OF PARTICLES
1.00-1.01	0.0
1.01-1.02	6.50E 03
1.02-1.03	2.08E 05
1.03-1.04	1.583
1.04-1.05	2.483
1.05-1.06	4.13E 07
1.06-1.07	1.441E 09
1.07-OVER	2.05E 09
TOTAL	0.0

TABLE 15

\*\*\*\*\*  
 \*\* ORBITAL FLUX STUDY WITH COMPOSITE SPECTRUM \*\*  
 \*\* UNIFORMITY FACTORS (UF) APPLIED FOR THIS RUN ARE: 1.73 \*\*  
 \*\* MAGNETIC COORDINATES 2 AND 3 COMPUTED BY INVAHA OF 1972 WITH ALL M.A. \*\*  
 \*\* FOR INFORMATION 3: EXPLANATION CONTACT L.C. STASSINGPULCULS AT NASA-GSFC CODE 631 \*\*  
 \*\*\*\*\*  
 \*\* UNIFORMITY FACTORS (UF) APPLIED FOR THIS RUN ARE: 1.73 \*\*  
 \*\* MAGNETIC COORDINATES 2 AND 3 COMPUTED BY INVAHA OF 1972 WITH ALL M.A. \*\*  
 \*\* FOR INFORMATION 3: EXPLANATION CONTACT L.C. STASSINGPULCULS AT NASA-GSFC CODE 631 \*\*  
 \*\*\*\*\*

\*\*\*\*\* COMPOSITE ORBIT SPECTRUM \*\*\*\*\* TABLE 5.0000 YH(S)

ENERGY RANGES >(MEV)	AVERAGED TOTAL FLUX #/CM <sup>2</sup> /SEC	AVERAGED TOTAL FLUX #/CM <sup>2</sup> /DAY	SPECTRUM PERCENT	ENERGY LEVELS >(MEV)	AVERAGED INTENS. FLUX #/CM <sup>2</sup> /SEC	AVERAGED INTENS. FLUX #/CM <sup>2</sup> /DAY	INTEGRAL FLUX #/CM <sup>2</sup> /TAU	INTEGRAL FLUX #/CM <sup>2</sup> /TAU	AVERAGED DIFFER. FLUX #/CM <sup>2</sup> /SEC/KEY
0.000-5.000	3.662E 07	3.176E 12	67.62%	0.0001-3.072 37	4.954E 13	4.356E 13	3.32E 13	3.32E 13	3.166E 05
1.000-1.500	5.452E 05	9.712E 06	1.31%	7.000E-314.376E 37	5.792E 12	5.522E 12	5.52E 12	1.076E 14	2.719E 05
1.500-2.000	1.124E 05	2.048E 06	0.31%	4.000E-303.676E 37	3.176E 12	3.076E 12	3.07E 12	6.255E 13	2.441E 05
2.000-3.000	2.824E 04	5.442E 05	0.75%	1.052E 07	1.472E 11	1.428E 11	1.42E 11	5.063E 13	1.344E 05
3.000-4.000	4.315E 03	8.131E 04	0.22%	5.000E	5.124E 10	4.924E 10	4.92E 10	3.121E 13	6.151E 04
4.000-5.000	6.030E 02	1.151E 04	0.32%	2.000E-200.000	7.048E 09	6.748E 09	6.74E 09	2.311E 13	2.363E 04
5.000-6.000	3.445E 01	6.576E 03	0.18%	5.000E	1.784E 10	1.724E 10	1.72E 10	1.711E 13	8.602E 03
6.000-OVER	1.051E 01	2.000E 02	0.28%	2.000E	5.032E 08	4.832E 08	4.83E 08	1.465E 13	3.595E 03
TOTAL	3.674E 07	3.176E 12	67.67%	0.0001-3.072 37	4.954E 13	4.356E 13	3.32E 13	3.32E 13	3.166E 05

\*\*\*\*\* EXPOSURE INDEX: ENERGY > 5.000 MEV \*\*\*\*\*

ENERGY RANGES >(MEV)	EXPOSURE DURATION (HOURS)	TOTAL # OF PARTICLES
1.0E-1.0E1	0.969	0.0
1.0E1-1.0E2	3.067	6.756E 02
1.0E2-1.0E3	0.167	2.526E 04
1.0E3-1.0E4	0.820	3.654E 05
1.0E4-1.0E5	4.283	1.759E 07
1.0E5-1.0E6	5.817	6.435E 08
1.0E6-1.0E7	7.917	1.540E 10
TOTAL	24.030	5.785E 10

TABLE 16

\*\*\*\*\*  
 ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VETTES AP4; AEG, A517 PJB SC1AB MAALIM \*\*\*\*\* UNIFLUX OF 1979  
 UNCERTAINTY FACTORS (UP) APPLIED FOR THIS RUN ARE: FOR PROTONS (APR) UP= 1.3; FOR INNER ZONE ELECTRONS (AEO) UP= 1.0  
 MAGNETIC COORDINATES Y AND L COMPUTED BY INVABA OF 1972 WITH ALL TAG MODEL 4; DRACRACLOUGH ET AL. 168 TRM 1375 & TIME= 1989.5  
 VEHICLE: WLLX3.9- MAX \*\* INCLINATION= 60DEG \*\* PERIGEE= 3489KM \*\* APOGEE= 3834KM \*\* B/L CRBIT TARD: TD5446 \*\* PERIOD= 2.876  
 FOR INFORMATION OR EXPLANATION CONTACT E.C. STASINPOULOS AT NASA-GSPCCODP 631, GREENBELT, MARYLAND 20771 TEL. (301)-344-6367  
 \*\*\*\*\*  
 ERIONS \*\*\*\*\*

\*\*\*\*\* SPECTRUM IN PERCENT DELTA ENERGY \*\*\*\*\* TAU= 5.0000 YEAR(S)

ENERGY RANGES > (MEV)	AVERAGED TOTAL FLUX #/CM**2/SEC	AVERAGED SPECTRUM TOTAL FLUX #/CM**2/DAY PERCENT	ENERGY LEVELS > (MEV)	AVERAGED INTEG.FLUX #/CM**2/SEC	AVERAGED INTEGR.FLUX #/CM**2/DAY	AVERAGED INTEG.FLUX #/CM**2/TAU	AVERAGED DIFFER.FLUX #/CM**2/SLC/KEY
1000-2.000	2.547E 06	2.201E 11	0.00E-014	0.091E 00	3.5337E 11	4.513E 14	2.596E 04
2.000-5.000	1.482E 05	1.281E 10	0.00E-013	3.99E 06	2.470E 11	5.360E 14	1.536E 04
5.000-10.000	7.726E 04	6.676E 09	1.00E 00	8.07E 05	5.745E 10	1.520E 14	1.534E 02
10.000-25.000	3.075E 04	2.657E 09	1.000	4.367E 05	3.546E 10	1.720E 14	4.816E 02
25.000-50.000	4.745E 03	4.099E 08	2.000	3.108E 05	2.212E 10	3.977E 13	5.269E 01
50.000-100.000	4.418E 03	3.817E 08	3.000	1.985E 05	1.713E 10	3.226E 13	5.267E 01
100.000-OVER	1.139E 01	9.841E 05	4.000	1.342E 05	1.405E 10	2.773E 13	2.773E 01
	1.019E 05	8.804E 05	5.000	1.342E 04	1.162E 09	2.134E 13	1.703E 01
	1.019E 05	8.804E 05	6.000	5.453E 04	7.375E 09	1.343E 13	1.170E 00
	1.019E 05	8.804E 05	7.000	2.339E 04	3.271E 09	3.688E 12	1.703E 00
	1.019E 05	8.804E 05	8.000	1.009E 04	1.219E 08	1.594E 12	3.334E 01
	1.019E 05	8.804E 05	10.000	7.126E 03	3.2157E 08	1.126E 12	5.268E 01
	1.019E 05	8.804E 05	20.000	5.049E 03	3.229E 08	1.126E 12	2.684E 01
	1.019E 05	8.804E 05	30.000	3.314E 03	3.229E 08	1.126E 12	1.584E 01
	1.019E 05	8.804E 05	40.000	3.092E 03	3.229E 08	1.126E 12	1.163E 01
	1.019E 05	8.804E 05	45.000	3.161E 03	2.733E 08	1.126E 12	1.266E 02
	1.019E 05	8.804E 05	50.000	2.776E 03	2.368E 08	1.126E 12	6.668E 02
	1.019E 05	8.804E 05	60.000	1.616E 03	1.368E 08	1.126E 12	3.586E 02
	1.019E 05	8.804E 05	80.000	1.152E 03	1.323E 08	1.126E 12	3.586E 02
	1.019E 05	8.804E 05	100.000	2.659E 02	1.323E 08	1.126E 12	7.815E 03
	1.019E 05	8.804E 05	200.000	1.205E 02	1.323E 08	1.126E 12	3.645E 03
	1.019E 05	8.804E 05	300.000	1.205E 01	1.323E 08	1.126E 12	1.323E 03
	1.019E 05	8.804E 05	350.000	5.304E 01	1.323E 08	1.126E 12	1.323E 03
	1.019E 05	8.804E 05	400.000	3.059E 01	1.323E 08	1.126E 12	3.323E 04
	1.019E 05	8.804E 05	500.000	1.019E 01	1.323E 08	1.126E 12	1.120E 04

TOTAL 2.858E 06 2.470E 11 69.368

\*\*\*\*\* EXPOSURE INDEX: ENERGY>5.000 MEV \*\*\*\*\*

INTENSITY RANGES #/CM**2/SEC	EXPOSURE DURATION (HOURS)	TOTAL PARTICLES	ACCUMULATED PARTICLES
ZERO FLUX	7.233	0	0
1.E0-1.E1	0.283	7.943E 03	7.943E 03
1.E1-1.E2	1.117	1.725E 05	1.725E 05
1.E2-1.E3	1.467	2.107E 06	2.107E 06
1.E3-1.E4	1.833	2.735E 07	2.735E 07
1.E4-1.E5	3.817	6.138E 08	6.138E 08
1.E5-1.E6	7.883	1.207E 10	1.207E 10
1.E6-1.E7	0.367	1.407E 09	1.407E 09
1.E7-OVER	0.0	0.0	0.0
TOTAL	24.000	1.405E 10	1.405E 10

TABLE 17

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVELOPMENTS: VETES APR; AEG, AE17 FOR SCALAR MAXIMUM ELECTRONS UNIPOLAR OF 1975  
 UNCERTAINTY FACTORS (UP) APPLIED FOR THIS RUN ARE: 1.0 FOR INNER ZONE ELECTRONS (AEG) - UP= 1.0  
 MAGNETIC COORDINATES (B) AND L-VALUE COMPUTED BY INVADG CODE 4: BANGACLUJH FT AL TCR-TRE 1 35 TIME= 1981.5  
 VEHICLE: MVLX3-S-MAX \*\* INCLINATION: 60 DEG \*\* PERIGEE= 389 KM \*\* APOGEE= 601 KM \*\* ORBIT TAPE: 1D546 PERIOD= 2.476  
 FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINOPOULOS AT NASA-GSFC CODE 601 GREENBELT, MARYLAND 20771 TEL. (301)-344-8067  
 \*\*\*\*\* ELECTRONS LC \*\*\*\*\*

\*\*\*\*\* SPECTRUM IN PERCENT DELTA ENERGY \*\*\*\*\*

ENERGY RANGES >(MEV)	AVERAGED TOTAL FLUX %/CM**2/SEC	AVERAGED TOTAL FLUX %/CM**2/DAY	SPECTRUM PERCENT
1000-1500	9.30E 07	5.44E 12	6.906
1500-2000	1.86E 06	1.60E 11	6.975
2000-3000	4.71E 04	4.07E 09	0.134
3000-4000	2.60E 04	2.24E 09	0.028
4000-5000	5.95E 03	5.14E 08	0.001
5000-6000	1.72E 01	9.93E 06	0.000
6000-CYB	2.92E -01	5.81E 04	0.000
TOTAL	6.50E 07	5.62E 12	69.100

\*\*\*\*\* EXPOSURE INDEX: ENERGY > 5000 MEV \*\*\*\*\*

INTENSITY RANGES %/CM**2/SEC	EXPOSURE DURATION (HOURS)	TOTAL # OF ACCUMULATED PARTICLES
ZINC FLUX	1.000	0.056E 04
1.00-1.25	0.033	5.28E 04
1.25-1.50	0.100	4.81E 05
1.50-1.75	0.300	4.81E 08
1.75-2.00	2.617	5.17E 10
2.00-2.25	10.717	2.05E 11
2.25-2.50	8.950	1.57E 11
1.25-OVER	24.000	1.78E 11
TOTAL		

\*\*\*\*\* COMPOSITE ORBIT SPECTRUM \*\*\*\*\* TAU= 5.0000 YR(S)

ENERGY LEVELS >(MEV)	AVERAGED TOTAL FLUX %/CM**2/SEC	AVERAGED TOTAL FLUX %/CM**2/DAY	AVERAGED INTEGRAL FLUENCE	OUTER ZONE	AVERAGE FLUX
4000E-019	8.13E 07	1.23E 12	1.46E 16	16	5.89E 05
4000E-017	8.82E 07	1.32E 12	1.41E 16	16	5.95E 05
4000E-015	2.96E 07	4.22E 11	4.07E 15	15	4.37E 05
3000	1.26E 06	1.09E 11	1.93E 15	15	1.08E 05
4000	5.00E 06	4.27E 11	4.91E 14	14	1.34E 05
5000	1.00E 06	8.78E 10	1.38E 14	14	6.87E 03
6000	5.87E 05	5.08E 10	7.52E 13	13	1.49E 03
7000	2.76E 05	2.38E 10	3.82E 13	13	5.36E 02
8000	1.28E 05	1.13E 10	1.66E 13	13	2.46E 02
1.250	8.03E 04	6.94E 09	9.91E 12	12	1.46E 01
1.750	3.17E 04	2.74E 09	3.70E 12	12	8.91E 01
2.250	2.32E 04	2.01E 09	2.82E 12	12	3.32E 01
2.750	1.04E 04	8.98E 08	1.27E 11	11	1.54E 01
3.250	4.47E 03	3.83E 08	5.29E 11	11	7.66E 00
3.750	1.70E 03	1.49E 08	2.01E 10	10	2.82E 00
4.250	8.85E 03	7.67E 08	1.03E 10	10	1.36E 00
4.750	2.85E 03	2.47E 08	3.39E 09	09	5.06E 00
5.250	1.21E 03	1.05E 08	1.51E 09	09	2.27E 00
5.750	7.58E 02	6.60E 07	4.49E 08	08	1.49E 00
6.250	3.92E 01	3.33E 06	2.86E 07	07	8.27E 00
6.750	2.32E 01	2.00E 06	1.66E 06	06	4.69E 00
7.250	0.00	0.00	0.00	00	0.00
7.500	0.00	0.00	0.00	00	0.00
TOTAL					

TABLE 18

\*\*\*\*\*  
 \*\* DRETAI FLUX STUDY WITH CONCENTRATED PARTICLE ENTRAINMENT \*\*  
 \*\* UNGRAVITY CHANGES (UP AND DOWN) IN THE FLUX ARE \*\*  
 \*\* MAGNETIC CHANGES IN THE FLUX ARE \*\*  
 \*\* FURTHER JOURNAL OF THE PLANETARY CONTACT LOGS \*\*  
 \*\*\*\*\*  
 \*\* DRETAI FLUX STUDY WITH CONCENTRATED PARTICLE ENTRAINMENT \*\*  
 \*\* UNGRAVITY CHANGES (UP AND DOWN) IN THE FLUX ARE \*\*  
 \*\* MAGNETIC CHANGES IN THE FLUX ARE \*\*  
 \*\* FURTHER JOURNAL OF THE PLANETARY CONTACT LOGS \*\*  
 \*\*\*\*\*

\*\*\*\*\* SPECTRUM IN PERCENT (ALTA ENERGY) \*\*\*\*\*

ENERGY RANGES >(MEV)	AVERAGE TOTAL FLUX #/CM <sup>2</sup> /SEC	AVERAGE TOTAL FLUX #/CM <sup>2</sup> /SEC	PERCENT	ERRORS
1000-2000	5.548E C6	4.750 C1	45.757	
2000-4000	4.450E C5	4.150 C10	5.752	
4000-6000	1.427E C5	1.400 C10	1.530	
6000-8000	9.854E C4	4.047E C5	0.550	
8000-10000	2.850E C3	4.170E C5	0.572	
10000-12000	2.850E C3	2.420E C6	0.000	
12000-14000	2.850E C3	5.000E C7	0.000	
14000-16000	2.850E C3	2.000E C7	0.000	
16000-18000	2.850E C3	2.000E C7	0.000	
18000-20000	2.850E C3	2.000E C7	0.000	
TOTAL	6.821E C6	5.425 C11	79.613	

\*\*\*\*\* REFUSURE INVERT ENERGY RANGE \*\*\*\*\*

RANGES	INTENSITY #/CM <sup>2</sup> /SEC (HOURS)	EXPOSURE CREATION	TOTAL # OF PARTICLES	ERRORS
1200-1400	6.400	1.000	6.400	
1400-1600	6.400	1.000	6.400	
1600-1800	6.400	1.000	6.400	
1800-2000	6.400	1.000	6.400	
2000-2200	6.400	1.000	6.400	
2200-2400	6.400	1.000	6.400	
2400-2600	6.400	1.000	6.400	
2600-2800	6.400	1.000	6.400	
2800-3000	6.400	1.000	6.400	
3000-3200	6.400	1.000	6.400	
3200-3400	6.400	1.000	6.400	
3400-3600	6.400	1.000	6.400	
3600-3800	6.400	1.000	6.400	
3800-4000	6.400	1.000	6.400	
TOTAL	24.000	2.000	48.000	

\*\*\*\*\* CONCENTRATED PARTICLE SPECTRUM \*\*\*\*\*

ENERGY RANGES >(MEV)	AV. FLUX #/CM <sup>2</sup> /SEC	AV. FLUX #/CM <sup>2</sup> /SEC	PERCENT	ERRORS
1000-2000	10.000E C6	10.000E C6	100.000	
2000-4000	1.000E C5	1.000E C5	10.000	
4000-6000	1.000E C5	1.000E C5	10.000	
6000-8000	1.000E C5	1.000E C5	10.000	
8000-10000	1.000E C5	1.000E C5	10.000	
10000-12000	1.000E C5	1.000E C5	10.000	
12000-14000	1.000E C5	1.000E C5	10.000	
14000-16000	1.000E C5	1.000E C5	10.000	
16000-18000	1.000E C5	1.000E C5	10.000	
18000-20000	1.000E C5	1.000E C5	10.000	
TOTAL	14.000E C6	14.000E C6	140.000	

TABLE 19





\*\*\*\*\* ORBITAL FLUX STUDY WITH COMBUSTIBLE PARTICLE ENVIRONMENTS: VEYES APRI, AEG, AE17, 119, 201, MAXIMUM \*\*\*\*\* UNIFLUX OF 1979 \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS BIN ARE: 1.00 FOR UNIFLUX, 1.00 FOR PARTICLE CONCENTRATIONS, 1.00 FOR TIME PERIODS \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES AND L COMPUTED BY INVARA OF 1972 WITH ALL MAGNETIC DATA FROM 1975 TO 1978 \*\*\*\*\*  
 \*\* VEHICLE INFORMATION: 30MTR, 30MTR, 30MTR, 30MTR, 30MTR, 30MTR, 30MTR, 30MTR, 30MTR, 30MTR \*\*\*\*\*  
 \*\* FOR INFORMATION ON EVALUATION CONTACT: STAFF/STAFF/STAFF/STAFF/STAFF/STAFF/STAFF/STAFF/STAFF/STAFF \*\*\*\*\*  
 \*\*\*\*\* FRITIKS \*\*\*\*\*

\*\*\*\*\* SPECTRUM IN PERCENT DELTA ENERGY \*\*\*\*\*

ENERGY RANGES > (MEV)	AVERAGE TOTAL FLUX #/CM <sup>2</sup> /SEC	AVERAGED TOTAL FLUX #/CM <sup>2</sup> /SEC	SPECTRUM PERCENT
1000-2000	1.140E 07	7.851E 11	76.79
2000-5000	4.311E 06	1.654E 10	6.47
5000-10000	1.721E 05	1.877E 10	1.07
10000-25000	3.762E 04	3.250E 09	0.23
25000-50000	1.517E 03	1.211E 08	0.09
50000-100000	1.574E 02	1.274E 07	0.01
100000-500000	5.671E 01	4.892E 06	0.00
500000-OVER	2.039E 01	1.761E 05	0.00
TOTAL	1.252E 07	1.002E 14	77.83

\*\*\*\*\* EXPOSURE INDEX: ENERGY > 0.00 MEV \*\*\*\*\*

INTENSITY RANGES	EXPOSURE CUMULATION	TOTAL # OF ACCUMULATE
1.00-1.01	7.777	2.0
1.01-1.02	1.000	1.0
1.02-1.03	1.354	2.0
1.03-1.04	1.750	2.0
1.04-1.05	2.147	4.0
1.05-1.06	2.544	4.0
1.06-1.07	2.941	4.0
1.07-OVER	3.338	4.0
TOTAL	24.000	2.117E 10

\*\*\*\*\* CONCRETE ENRIT SPECTRUM \*\*\*\*\* TIME 100000 YEARS \*\*\*\*\*

ENERGY LEVELS > (MEV)	AVERAGED INTFC FLUX #/CM <sup>2</sup> /SEC	AVERAGED INTFC FLUX #/CM <sup>2</sup> /SEC	AVERAGED INTFC FLUX #/CM <sup>2</sup> /SEC	AVERAGED INTFC FLUX #/CM <sup>2</sup> /SEC
1000-2000	1.140E 07	1.140E 07	2.534E 12	4.815E 04
2000-5000	4.311E 06	1.312E 12	1.679E 12	2.858E 04
5000-10000	1.721E 05	4.102E 11	7.414E 12	4.737E 04
10000-25000	3.762E 04	2.138E 11	3.502E 14	1.002E 04
25000-50000	1.517E 03	5.203E 10	1.723E 13	2.456E 03
50000-100000	1.574E 02	3.472E 10	6.237E 13	2.037E 03
100000-500000	5.671E 01	1.102E 10	3.379E 13	1.166E 02
500000-OVER	2.039E 01	4.304E 09	1.102E 13	1.469E 01
1000-2000	1.140E 07	3.034E 09	1.102E 13	1.143E 01
2000-5000	4.311E 06	2.107E 08	1.102E 13	2.835E 01
5000-10000	1.721E 05	1.530E 08	1.102E 13	4.937E 01
10000-25000	3.762E 04	4.044E 07	1.102E 13	1.769E 02
25000-50000	1.517E 03	5.822E 07	1.102E 13	4.032E 02
50000-100000	1.574E 02	4.201E 07	7.477E 12	2.112E 02
100000-500000	5.671E 01	2.104E 07	3.522E 12	1.225E 02
500000-OVER	2.039E 01	1.615E 07	3.522E 12	1.481E 02
1000-2000	1.140E 07	4.832E 06	1.102E 13	2.813E 02
2000-5000	4.311E 06	3.844E 06	1.102E 13	1.488E 02
5000-10000	1.721E 05	2.265E 06	1.102E 13	1.432E 02
10000-25000	3.762E 04	2.265E 06	1.102E 13	1.432E 02
25000-50000	1.517E 03	1.517E 03	1.102E 13	2.450E 02
50000-100000	1.574E 02	1.574E 02	1.102E 13	1.177E 02
100000-500000	5.671E 01	2.720E 01	1.102E 13	2.503E 02
500000-OVER	2.039E 01	1.676E 01	1.102E 13	2.938E 02

TABLE 21

\*\*\*\*\*  
 \*\* DATA FILE STUDY WITH APPLIED PASSIVE ENVIRONMENTAL MONITORING SYSTEMS. METERS ARE 1.00 METER MAXIMUM UNIFLUX OF 1979 \*\*  
 \*\* UNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: FOR PROTONS (A) 1.00, FOR ELECTRONS (A) 1.00, FOR GAMMA (A) 1.00 \*\*  
 \*\* MAGNETIC COORDINATES (X) AND (Y) COMPUTED BY INVARA (E 1972) WITH ALL MAG. MULTIPLIER 4.15, BOUNDARY TIME 149-304 1975, TIME 1989.5 \*\*  
 \*\* MESSAGE NUMBER 632984 \*\* \* PERIOD 3.6588 \*\*  
 \*\* FOR INFORMATION ON EXPLANATION CONTACT C. G. STANFIELD AT NASA-SCIENCE APPLICATIONS CENTER, 7000 MAINTENANCE ROAD, WASHINGTON 20771. (301) 344-8067 \*\*  
 \*\*\*\*\*  
 \*\* ELECTRONIC \*\*\*\*\*  
 \*\*\*\*\*

\*\*\*\*\* SPECTRUM IN PERCENT DELTA ENERGY \*\*\*\*\*

ENERGY RANGES > (MEV)	AVERAGE TOTAL FLUX #/CM**2/SEC	AVERAGE TOTAL FLUX #/CM**2/SEC	SPECTRUM PERCENT
1.000-1.500	6.433E 07	4.944E 10	48.608
1.500-2.000	1.540E 06	1.365E 11	1.921
2.000-2.500	1.269E 06	1.133E 10	3.143
2.500-3.000	3.272E 04	4.581E 09	3.073
3.000-4.000	2.558E 04	3.213E 09	6.012
4.000-5.000	1.925E 04	4.853E 08	3.303
5.000-6.000	2.327E 02	2.333E 08	6.000
6.000-6.500	1.757E 02	1.165E 07	6.000
6.500-7.000	1.541E 01	5.436E 06	6.000
7.000-8.000	3.012E 05	3.121E 10	6.000
8.000-9.000	2.414E 05	2.413E 10	6.000
9.000-10.000	1.434E 05	1.239E 10	6.000
10.000-11.000	5.391E 04	7.390E 09	6.000
11.000-12.000	3.329E 04	4.509E 09	6.000
12.000-13.000	2.843E 04	2.491E 09	6.000
13.000-14.000	2.148E 04	1.337E 09	6.000
14.000-15.000	1.627E 04	1.432E 09	6.000
15.000-16.000	1.271E 04	1.094E 09	6.000
16.000-17.000	8.411E 03	7.267E 08	6.000
17.000-18.000	5.575E 03	4.917E 08	6.000
18.000-19.000	3.750E 03	3.217E 08	6.000
19.000-20.000	2.459E 03	2.175E 08	6.000
20.000-21.000	3.765E 02	4.903E 07	6.000
21.000-22.000	1.359E 02	1.174E 07	6.000
22.000-23.000	1.023E 01	8.935E 06	6.000
23.000-24.000	6.248E -01	5.430E 06	6.000
24.000-25.000	9.000E -01	9.000E 05	6.000

TOTAL 5.583E 07 4.944E 10 67.143

\*\*\*\*\* EXPOSURE INDEX ENERGY 4800 MUV \*\*\*\*\*

RANGES #/CM**2/SEC (MURS)	EXPOSURE	TOTAL # OF ACCUMULATED PARTICLES
1.E0-1.E1	1.587	0.0
1.E1-1.E2	0.100	1.622E 05
1.E2-1.E3	0.243	5.422E 04
1.E3-1.E4	0.367	5.528E 05
1.E4-1.E5	0.383	5.428E 06
1.E5-1.E6	0.600	7.284E 07
1.E6-1.E7	7.653	1.557E 10
1.E7-TOTAL	0.0	0.0

TOTAL 29.022 1.566E 11

TABLE 222

\*\*\*\*\*  
 \*\* JINCO AL PLIX... \*\*  
 \*\* JINCO... \*\*  
 \*\* MECH... \*\*  
 \*\* ELECT... \*\*  
 \*\*\*\*\*

ENERGY EARTH MAGN	PYR... W/... MAGN	THERM... W/... MAGN	SPECTRUM MAGN	ENERGY (KWH)	AVG... /... MAGN	AVERAGE TEMPERATURE (C)	SURFACE AREA (SQ. FT.)	AVG. RADIATION (BTU/SQ. FT./HR.)	VOLUME (CUBIC FT.)
1100-1200	1200-1300	1300-1400	1400-1500	1500-1600	1600-1700	1700-1800	1800-1900	1900-2000	2000-2100
...	...	...	...	...	...	...	...	...	...
TOTAL									
*****									
NET... ..									
GROSS... ..									
TOTAL... ..									

TABLE 23

\*\*\*\*\*  
 \*\* CREDIT FLUX ENERGY RANGE 1.000 TO 5.000 MEV (1.000 TO 5.000 MEV) \*\*  
 \*\* SOURCE: 1.000 TO 5.000 MEV (1.000 TO 5.000 MEV) \*\*  
 \*\* DATE: 1979 (1979) \*\*  
 \*\* INSTRUMENT: 1.000 TO 5.000 MEV (1.000 TO 5.000 MEV) \*\*  
 \*\* ENERGY RANGE: 1.000 TO 5.000 MEV (1.000 TO 5.000 MEV) \*\*  
 \*\* INTEGRATION: 1.000 TO 5.000 MEV (1.000 TO 5.000 MEV) \*\*  
 \*\* CORRECTIONS: 1.000 TO 5.000 MEV (1.000 TO 5.000 MEV) \*\*  
 \*\*\*\*\*

ENERGY RANGE (MEV)	COUNTS	INTENSITY (C/HR)	AVERAGE INTENSITY (C/HR)	AVG. FLUX (C/HR)	AVG. FLUX (C/HR)	AVERAGE FLUX (C/HR)	AVG. FLUX (C/HR)	AVG. FLUX (C/HR)	AVG. FLUX (C/HR)	AVG. FLUX (C/HR)	AVG. FLUX (C/HR)	AVG. FLUX (C/HR)	AVG. FLUX (C/HR)
1.000-1.250	14	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000	1.000
1.250-1.500	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
1.500-1.750	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
1.750-2.000	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
2.000-2.250	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
2.250-2.500	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
2.500-2.750	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
2.750-3.000	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
3.000-3.250	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
3.250-3.500	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
3.500-3.750	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
3.750-4.000	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
4.000-4.250	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
4.250-4.500	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
4.500-4.750	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
4.750-5.000	15	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125	1.125
TOTAL	140	10.000	10.000	10.000	10.000	10.000	10.000	10.000	10.000	10.000	10.000	10.000	10.000
EXP. SURT INSERT FOR > 5.000 MEV													
INTENSITY EXP. SURT INSERT FOR > 5.000 MEV													
AVG. FLUX (C/HR)													
7500 FLUX	7500	7500	7500	7500	7500	7500	7500	7500	7500	7500	7500	7500	7500
1.500-1.750	1.500	1.500	1.500	1.500	1.500	1.500	1.500	1.500	1.500	1.500	1.500	1.500	1.500
1.750-2.000	1.750	1.750	1.750	1.750	1.750	1.750	1.750	1.750	1.750	1.750	1.750	1.750	1.750
2.000-2.250	2.000	2.000	2.000	2.000	2.000	2.000	2.000	2.000	2.000	2.000	2.000	2.000	2.000
2.250-2.500	2.250	2.250	2.250	2.250	2.250	2.250	2.250	2.250	2.250	2.250	2.250	2.250	2.250
2.500-2.750	2.500	2.500	2.500	2.500	2.500	2.500	2.500	2.500	2.500	2.500	2.500	2.500	2.500
2.750-3.000	2.750	2.750	2.750	2.750	2.750	2.750	2.750	2.750	2.750	2.750	2.750	2.750	2.750
3.000-3.250	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000
3.250-3.500	3.250	3.250	3.250	3.250	3.250	3.250	3.250	3.250	3.250	3.250	3.250	3.250	3.250
3.500-3.750	3.500	3.500	3.500	3.500	3.500	3.500	3.500	3.500	3.500	3.500	3.500	3.500	3.500
3.750-4.000	3.750	3.750	3.750	3.750	3.750	3.750	3.750	3.750	3.750	3.750	3.750	3.750	3.750
4.000-4.250	4.000	4.000	4.000	4.000	4.000	4.000	4.000	4.000	4.000	4.000	4.000	4.000	4.000
4.250-4.500	4.250	4.250	4.250	4.250	4.250	4.250	4.250	4.250	4.250	4.250	4.250	4.250	4.250
4.500-4.750	4.500	4.500	4.500	4.500	4.500	4.500	4.500	4.500	4.500	4.500	4.500	4.500	4.500
4.750-5.000	4.750	4.750	4.750	4.750	4.750	4.750	4.750	4.750	4.750	4.750	4.750	4.750	4.750
TOTAL	40.000	40.000	40.000	40.000	40.000	40.000	40.000	40.000	40.000	40.000	40.000	40.000	40.000

TABLE 24

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 \*\* ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VECTORS APPLIED FOR ALL 7 FOR MAXIMUM FLUX OF 1979  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PRACTICE (APP) - UF = 1.0; FOR INSTRUMENT ELECTRONICS (AFC) - UF = 1.3  
 \*\* MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALL MAGNETIC MODEL 4; MAGNETIC COORDINATES L, B, TIME = 1989.5  
 \*\* VEHICLE - NUKLIS-MAX AT INCLINATION - 30.2 DEGREES, ALTITUDE - 166.7 KM, SPEED - 7.7 KM/SEC, PERIOD - 105.4 MIN, MASS - 1.0 KG  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT E. J. STASSINGFOLG AT NASA-JSC CODE 621, BUILDING 31, WALKER T. MARSHALL, P.O. BOX 2171, LIAISON - 344-8067  
 \*\*\*\*\*

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM <sup>2</sup> /SEC	POSITION AT LONGITUDE (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	VELOCITY (M/SEC)	PERIOD (MIN)	PERIOD (HOURS)	PERIOD (DAYS)	PERIOD (YEARS)	PERIOD (CENTURIES)	PERIOD (MILLENNIA)	PERIOD (BILLIONS)	PERIOD (TRILLIONS)
1	2.242E 04	-105.698	-14.14	160.873	1.17000	0.18473	0.18473	1.01	4.247E 07				
2	2.4505E 04	-114.97	-18.42	160.873	3.45066	0.18473	0.18473	1.01	4.647E 07				
3	5.055E 04	-22.515	-24.40	160.873	4.91667	0.18473	0.18473	1.01	6.827E 07				
4	6.828E 04	-49.359	-18.70	160.873	7.89999	0.18473	0.18473	1.01	7.545E 07				
5	8.500E 04	-72.238	-14.96	160.873	9.36666	0.18473	0.18473	1.01	8.273E 07				
6	2.156E 04	-104.769	-13.85	160.873	11.49777	0.18473	0.18473	1.01	9.321E 07				
7	1.158E 04	-97.422	4.31	150.814	11.49777	0.18473	0.18473	1.01	1.630E 07				
8	1.772E 04	56.526	-3.97	155.433	14.95667	0.18473	0.18473	1.01	2.631E 07				
9	2.803E 04	13.543	-17.68	160.873	17.39999	0.18473	0.18473	1.01	3.863E 07				
10	4.655E 04	-23.076	-19.68	155.473	18.99999	0.18473	0.18473	1.01	5.559E 07				
11	6.722E 04	-46.922	-17.33	160.873	21.33333	0.18473	0.18473	1.01	6.666E 07				
12	4.724E 04	-74.954	-20.59	160.873	23.99999	0.18473	0.18473	1.01	5.822E 07				

\*\*\*\*\*  
 \*\* TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY 25.000 MEV  
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\*\*\*\*\*  
 \*\* ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VECTORS APPLIED FOR ALL 7 FOR MAXIMUM FLUX OF 1979  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PRACTICE (APP) - UF = 1.0; FOR INSTRUMENT ELECTRONICS (AFC) - UF = 1.0  
 \*\* MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALL MAGNETIC MODEL 4; MAGNETIC COORDINATES L, B, TIME = 1989.5  
 \*\* VEHICLE - NUKLIS-MAX AT INCLINATION - 30.2 DEGREES, ALTITUDE - 166.7 KM, SPEED - 7.7 KM/SEC, PERIOD - 105.4 MIN, MASS - 1.0 KG  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT E. J. STASSINGFOLG AT NASA-JSC CODE 621, BUILDING 31, WALKER T. MARSHALL, P.O. BOX 2171, LIAISON - 344-8067  
 \*\*\*\*\*

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM <sup>2</sup> /SEC	POSITION AT LONGITUDE (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	VELOCITY (M/SEC)	PERIOD (MIN)	PERIOD (HOURS)	PERIOD (DAYS)	PERIOD (YEARS)	PERIOD (CENTURIES)	PERIOD (MILLENNIA)	PERIOD (BILLIONS)	PERIOD (TRILLIONS)
1	1.921E 04	-105.923	-24.32	160.873	1.19067	0.18473	0.18473	1.01	2.184E 07				
2	2.027E 04	-15.775	-19.71	160.873	2.81814	0.18473	0.18473	1.01	1.830E 07				
3	2.827E 04	-41.794	-25.24	160.873	3.84333	0.18473	0.18473	1.01	3.561E 07				
4	5.931E 04	-53.074	-25.24	160.873	7.30000	0.18473	0.18473	1.01	3.264E 07				
5	2.242E 04	-68.928	-24.48	160.873	9.36666	0.18473	0.18473	1.01	2.531E 07				
6	1.649E 04	-67.422	4.31	160.873	11.49777	0.18473	0.18473	1.01	1.674E 07				
7	1.920E 04	-49.359	-8.38	160.873	11.49777	0.18473	0.18473	1.01	1.944E 07				
8	1.791E 04	-41.794	-8.83	160.873	14.95667	0.18473	0.18473	1.01	3.500E 07				
9	2.137E 04	-14.540	-19.68	160.873	17.39999	0.18473	0.18473	1.01	2.715E 07				
10	2.761E 04	-46.922	-19.68	160.873	18.99999	0.18473	0.18473	1.01	3.277E 07				
11	4.007E 04	-46.922	-17.33	160.873	21.33333	0.18473	0.18473	1.01	4.666E 07				
12	2.833E 04	-73.624	-20.59	160.873	23.99999	0.18473	0.18473	1.01	3.961E 07				

\*\*\*\*\*  
 \*\* TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY 25.000 MEV  
 \*\*\*\*\*

TABLE 25

\*\*\*\*\* ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VLTIC APOL AEOI AB17 LUP SAILAP MAXIMUM UNIFLUX OF 1979 \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PARTICLES (APR) - UFE = 1.0; FOR INTR ZONE ELECTRONS (AEK) - UFE = 1.0 \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES (L AND L) COMPUTED BY INVARA OF 1972 WITH ALLMAU. MODEL 4: BARRAGUET JCH ET AL. 1974-TRM 1975 \* TIME = 1989.5 \*\*\*\*\*  
 \*\* VEHICLE: MEX2-SMAX. AN INCLINATION CORRECTED TO EQUATORIAL PLANE. \*\*\*\*\*  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT L. G. STASSINOPOLIS AT NASA-JSC CODE 631, GREENBAY, WISCONSIN 53001 \*\*\*\*\*  
 \*\* TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 0.001 MeV \*\*\*\*\*

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM**2/SEC	POSITION AT LONGITUDE (DEG)	POSITION AT LATITUDE (DEG)	HEIGHT (KM)	PERIOD (HOURS)	FLUXES (CAUSE)	LINE (E. W.)	TOTAL FLUX #/CM**2/CH-RBT
1	1.524E 05	-110.840	-12.36	2580.65	1.26667	3.13917	1.43	3.335E 04
2	2.084E 05	84.24	-3.59	2795.77	4.66667	3.13917	1.50	7.726E 04
3	3.353E 05	-28.675	-7.59	2593.17	6.93333	3.13917	1.50	5.009E 04
4	3.182E 05	-68.455	-15.95	2593.17	7.26667	3.13917	1.45	4.510E 04
5	1.780E 05	-133.692	-11.14	2583.45	14.53333	3.13917	1.44	2.634E 04
6	1.244E 05	24.297	-4.13	2593.15	14.53333	3.13917	1.41	2.149E 04
7	1.551E 05	-2.863	-2.82	2585.41	17.23333	3.13917	1.43	2.673E 04
8	2.420E 05	-36.014	-9.24	2585.45	17.23333	3.13917	1.50	4.502E 04
9	3.494E 05	-64.824	-9.04	2585.43	19.61667	3.13917	1.50	4.321E 04
10	3.230E 05	-64.824	-14.26	2587.95	22.41333	3.13917	1.50	3.551E 04

\*\*\*\*\* ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VLTIC APOL AEOI AB17 LUP SAILAP MAXIMUM UNIFLUX OF 1979 \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PARTICLES (APR) - UFE = 1.0; FOR INTR ZONE ELECTRONS (AEK) - UFE = 1.0 \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES (L AND L) COMPUTED BY INVARA OF 1972 WITH ALLMAU. MODEL 4: BARRAGUET JCH ET AL. 1974-TRM 1975 \* TIME = 1989.5 \*\*\*\*\*  
 \*\* VEHICLE: MEX2-SMAX. AN INCLINATION CORRECTED TO EQUATORIAL PLANE. \*\*\*\*\*  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT L. G. STASSINOPOLIS AT NASA-JSC CODE 631, GREENBAY, WISCONSIN 53001 \*\*\*\*\*  
 \*\* TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 0.001 MeV \*\*\*\*\*

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM**2/SEC	POSITION AT LONGITUDE (DEG)	POSITION AT LATITUDE (DEG)	HEIGHT (KM)	PERIOD (HOURS)	FLUXES (CAUSE)	LINE (E. W.)	TOTAL FLUX #/CM**2/CH-RBT
1	3.97E 06	-106.609	-19.74	2687.05	1.30300	3.13917	1.45	4.004E 04
2	4.45E 06	-26.434	-3.59	2922.97	4.66667	3.13917	1.50	5.553E 04
3	5.38E 06	-69.521	-7.79	2923.36	7.26667	3.13917	1.52	1.022E 05
4	5.21E 06	-15.795	-15.95	2924.16	7.26667	3.13917	1.45	1.141E 05
5	4.14E 06	-133.692	-11.14	2583.45	14.53333	3.13917	1.45	4.432E 04
6	3.47E 06	-133.692	-11.14	2593.43	14.53333	3.13917	1.42	7.544E 04
7	4.03E 06	-31.764	-2.82	2925.77	17.23333	3.13917	1.41	4.111E 04
8	4.72E 06	-36.014	-9.04	2925.77	17.23333	3.13917	1.49	4.744E 04
9	4.42E 06	-64.824	-9.04	2925.77	19.61667	3.13917	1.50	4.544E 04
10	5.26E 06	-64.824	-14.26	2927.95	22.41333	3.13917	1.50	1.334E 05

TABLE 26

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VATTES APOLLO 860, ABET PUB SOLAR MAXIMUM UNIFLUX OF 1979  
 UNCERTAINTY FACTOR (UF) APPLIED FOR THIS RUN ARE: FOR PROBLEMS (APP) - JFE = 1.3, FOR LINEAR SCHEM EFFECTS (AS6) - IIR = 1.0  
 MAGNETIC COORDINATES (B AND L) COMPUTED BY ANNA FA OF 1972 WITH SLM, 1980 WITH JARISCOLOGUE FOR ALL TRACKS FROM 1979 - IIR = 1989, 5  
 VEHICLE: NULX3-SAX, INCLINATION = 60 DEG, PERIGEE = 390 KM, APOGEE = 383 KM, BZ1.0, IIR = 10974, IIR = 10974, IIR = 10974, IIR = 10974, IIR = 10974  
 FOR INFORMATION OR EXPLANATION CONTACT E. C. STANISLAPOULOS AT NASA-SPOC BOX 6711, GREENBELT, MARYLAND 20771, TEL. (301) 344-8067

TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 500 MEV

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM**2/SEC	POSITION AT WHICH ENCOUNTERED LONGITUDE (DEG)	POSITION AT WHICH ENCOUNTERED LATITUDE (DEG)	ENCLINATED ALTITUDE (KM)	CABIT TIME (HOURS)	FIELD (GAUSS)	LINE (L) (E.R.)	TOTAL FLUX PER ORBIT #/CM**2/GEHIT
1	7.31E 05	-116.921	-9.64	3883.25	1.71997	0.07170	1.73	1.73E 09
2	1.01E 06	-8.354	-11.39	3882.94	9.73334	0.06554	1.53	2.09E 09
3	1.13E 06	-97.034	-1.17	3887.91	19.51637	0.06353	1.70	4.25E 09
4	9.67E 05	-156.930	-2.15	3889.07	11.33333	0.06951	1.93	1.23E 09
5	9.77E 05	-20.726	-9.58	3882.99	15.76667	0.06432	1.91	1.23E 09
6	1.08E 06	-18.784	-13.91	3887.17	19.71666	0.06350	1.92	1.23E 09
8	1.12E 06	-57.750	-13.91	3883.36	21.53333	0.06350	1.70	1.73E 09

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VATTES APOLLO 860, ABET PUB SOLAR MAXIMUM UNIFLUX OF 1979  
 UNCERTAINTY FACTOR (UF) APPLIED FOR THIS RUN ARE: FOR PROBLEMS (APP) - JFE = 1.3, FOR LINEAR SCHEM EFFECTS (AS6) - IIR = 1.0  
 MAGNETIC COORDINATES (B AND L) COMPUTED BY ANNA FA OF 1972 WITH SLM, 1980 WITH JARISCOLOGUE FOR ALL TRACKS FROM 1979 - IIR = 1989, 5  
 VEHICLE: NULX3-SAX, INCLINATION = 60 DEG, PERIGEE = 390 KM, APOGEE = 383 KM, BZ1.0, IIR = 10974, IIR = 10974, IIR = 10974, IIR = 10974, IIR = 10974  
 FOR INFORMATION OR EXPLANATION CONTACT E. C. STANISLAPOULOS AT NASA-SPOC BOX 6711, GREENBELT, MARYLAND 20771, TEL. (301) 344-8067

TABLE OF PEAK AND TOTAL FLUXES PER PERIOD: ENERGY > 500 MEV

PERIOD NUMBER	PEAK FLUX ENCOUNTERED #/CM**2/SEC	POSITION AT WHICH ENCOUNTERED LONGITUDE (DEG)	POSITION AT WHICH ENCOUNTERED LATITUDE (DEG)	ENCLINATED ALTITUDE (KM)	CABIT TIME (HOURS)	FIELD (GAUSS)	LINE (L) (E.R.)	TOTAL FLUX PER ORBIT #/CM**2/GEHIT
1	7.32E 06	-116.821	-8.64	4883.25	1.71667	0.07270	1.63	4.38E 10
2	9.21E 06	-57.264	-13.91	4888.93	1.71666	0.06350	1.64	2.64E 10
4	9.31E 06	-97.034	-1.17	4889.07	11.33333	0.06951	1.63	2.64E 10
5	9.69E 06	-156.930	-2.15	4889.07	11.33333	0.06951	1.63	2.64E 10
6	9.78E 06	-20.726	-9.58	4882.99	15.76667	0.06432	1.65	2.64E 10
7	9.92E 06	-18.784	-13.91	4887.17	11.71666	0.06432	1.70	2.64E 10
8	9.17E 06	-57.750	-13.91	4883.36	21.68333	0.06350	1.70	2.64E 10

TABLE 27

\*\* DREITAL FLUX STUDY WITH COMPOSITE PARTICLE ENHANCER: THIS IS THE FIRST REPORT ON THE DREITAL FLUX STUDY. THE DREITAL FLUX STUDY IS A  
 \*\* UNCERTAINTY FACTORS (UP) AND (DN) FOR THE RUN AREA. THE RUN AREA IS THE AREA WHERE THE PARTICLES ARE DETECTED. THE DREITAL FLUX STUDY IS A  
 \*\* MAGNETIC COORDINATES IN AN INCLINATION OF 1972 WITH ANOMALOUS AGING. THE DREITAL FLUX STUDY IS A MAGNETIC COORDINATES IN AN INCLINATION  
 \*\* MAGNETIC COORDINATES IN AN INCLINATION OF 1972 WITH ANOMALOUS AGING. THE DREITAL FLUX STUDY IS A MAGNETIC COORDINATES IN AN INCLINATION  
 \*\* FOR INFORMATION OR EXPLANATION OF CONTACT INFORMATION, CONTACT INFORMATION AT NASA-SCSC, OFFICE OF THE DIRECTOR, NASA-SCSC, WASHINGTON, D.C. 20546-0001

PERIOD NUMBER	PEAK FLUX COUNTS (250000-250000)	POSITION AT LONGITUDE (C)	POSITION AT LATITUDE (DEG)	HEIGHT (KM)	UNIT TIME (HOURS)	FLUX (C)	FLUX (C)
1	10791 06	-125.412	-5.03	3400.00	17.7333	0.0124	0.0124
2	10818 06	-125.426	-5.03	3400.00	17.7333	0.0124	0.0124
3	10821 06	-125.427	-5.03	3400.00	17.7333	0.0124	0.0124
4	10741 06	-125.418	-5.03	3400.00	17.7333	0.0124	0.0124
5	10741 06	-125.418	-5.03	3400.00	17.7333	0.0124	0.0124
6	10821 06	-125.426	-5.03	3400.00	17.7333	0.0124	0.0124

\*\* DREITAL FLUX STUDY WITH COMPOSITE PARTICLE ENHANCER: THIS IS THE FIRST REPORT ON THE DREITAL FLUX STUDY. THE DREITAL FLUX STUDY IS A  
 \*\* UNCERTAINTY FACTORS (UP) AND (DN) FOR THE RUN AREA. THE RUN AREA IS THE AREA WHERE THE PARTICLES ARE DETECTED. THE DREITAL FLUX STUDY IS A  
 \*\* MAGNETIC COORDINATES IN AN INCLINATION OF 1972 WITH ANOMALOUS AGING. THE DREITAL FLUX STUDY IS A MAGNETIC COORDINATES IN AN INCLINATION  
 \*\* MAGNETIC COORDINATES IN AN INCLINATION OF 1972 WITH ANOMALOUS AGING. THE DREITAL FLUX STUDY IS A MAGNETIC COORDINATES IN AN INCLINATION  
 \*\* FOR INFORMATION OR EXPLANATION OF CONTACT INFORMATION, CONTACT INFORMATION AT NASA-SCSC, OFFICE OF THE DIRECTOR, NASA-SCSC, WASHINGTON, D.C. 20546-0001

PERIOD NUMBER	PEAK FLUX COUNTS (250000-250000)	POSITION AT LONGITUDE (C)	POSITION AT LATITUDE (DEG)	HEIGHT (KM)	UNIT TIME (HOURS)	FLUX (C)	FLUX (C)
1	10698 07	-125.412	-5.03	3400.00	17.7333	0.0124	0.0124
2	10741 07	-125.426	-5.03	3400.00	17.7333	0.0124	0.0124
3	10741 07	-125.427	-5.03	3400.00	17.7333	0.0124	0.0124
4	10741 07	-125.418	-5.03	3400.00	17.7333	0.0124	0.0124
5	10741 07	-125.418	-5.03	3400.00	17.7333	0.0124	0.0124
6	10821 07	-125.426	-5.03	3400.00	17.7333	0.0124	0.0124

TABLE 28



\*\*\*\*\* ORBITAL FLUX STUDY WITH POSSIBLE PARTICLE ENRICHMENTS: VETTES A60, A617 FOR SCLAR MAXIMUM \*\*\*\*\* UNIFLUX CF 1979 \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS OF 10% APPLIED TO THE SCLAR AREA FOR THIS ORBITAL FLUX STUDY \*\*\*\*\* UNIFLUX CF 1979 \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES 3 AND 4 CORRECTED FOR THE 1972 WITH ALL MAGNETIC FIELD DATA \*\*\*\*\* UNIFLUX CF 1979 \*\*\*\*\*  
 \*\* FOR INFORMATION OF EXPLANATION CONTACT 3066 AT NASA-CSCFC, SCLAR, CHARLOTTE, MARYLAND 20771, TEL. (301)-344-8067 \*\*\*\*\*

NUMBER	PERIOD 270**27/SEC	POSITION AT WHICH ENCOUNTED CONJUNCTION (COS)	LATITUDE (DEG)	ALTITUDE (KM)	CREDIT TIME (HOURS)	FILED(S) (CAUSE)	LINEID (REP.)	TOTAL FLUX PER ORBIT #7CM**2/CREDIT
1	8.432E CC	64.833	9.11	6389.46	0.11667	3.54200	1.57	3.215E 10
2	8.432E CC	168.658	7.50	6384.34	5.43333	3.04575	1.57	3.174E 10
3	8.432E CC	167.624	9.02	6384.51	9.41500	3.04575	1.56	2.194E 10
4	8.432E CC	-154.538	3.42	6389.03	15.49500	3.04575	1.55	2.039E 10
5	8.432E CC	-154.437	1.72	6389.03	15.49667	3.04575	1.55	1.535E 10
6	8.432E CC	166.507	9.11	6389.46	20.00000	3.04575	1.56	2.495E 10

\*\*\*\*\* ORBITAL FLUX STUDY WITH POSSIBLE PARTICLE ENRICHMENTS: VETTES A60, A617 FOR SCLAR MAXIMUM \*\*\*\*\* UNIFLUX CF 1979 \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS OF 10% APPLIED TO THE SCLAR AREA FOR THIS ORBITAL FLUX STUDY \*\*\*\*\* UNIFLUX CF 1979 \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES 3 AND 4 CORRECTED FOR THE 1972 WITH ALL MAGNETIC FIELD DATA \*\*\*\*\* UNIFLUX CF 1979 \*\*\*\*\*  
 \*\* FOR INFORMATION OF EXPLANATION CONTACT 3066 AT NASA-CSCFC, SCLAR, CHARLOTTE, MARYLAND 20771, TEL. (301)-344-8067 \*\*\*\*\*

NUMBER	PERIOD 270**27/SEC	POSITION AT WHICH ENCOUNTED CONJUNCTION (COS)	LATITUDE (DEG)	ALTITUDE (KM)	CREDIT TIME (HOURS)	FILED(S) (CAUSE)	LINEID (REP.)	TOTAL FLUX PER ORBIT #7CM**2/CREDIT
1	1.014E CC	140.023	17.11	6389.46	0.11667	3.04575	1.57	4.071E 09
2	1.014E CC	138.047	17.00	6384.34	5.43333	3.04575	1.57	4.412E 09
3	1.014E CC	136.071	15.31	6384.51	9.41500	3.04575	1.56	3.519E 09
4	1.014E CC	-134.095	15.42	6389.03	15.49500	3.04575	1.55	2.766E 09
5	1.014E CC	-134.095	15.42	6389.03	15.49667	3.04575	1.55	2.663E 09
6	1.014E CC	132.119	17.11	6389.46	20.00000	3.04575	1.56	3.351E 09

TABLE 29

\*\*\*\*\* ORBITA FLUX STUDY WITH AMPERSIE PARTIAL ENVIRONMENTAL VECTORS FROM 1972 TO 1979 \*\*\*\*\*  
 \*\* JNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: 1.00 FOR ALL DATA, 1.00 FOR ALL DATA, 1.00 FOR ALL DATA \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES (U) AND L COMPUTED BY INAVIA OF 1972 WITH ALL DATA, 1.00 FOR ALL DATA, 1.00 FOR ALL DATA \*\*\*\*\*  
 \*\* VECTORS (U) AND L COMPUTED BY INAVIA OF 1972 WITH ALL DATA, 1.00 FOR ALL DATA, 1.00 FOR ALL DATA \*\*\*\*\*  
 \*\* FOR INFORMATION ON EXPLANATION CONTACT STASINCEFOULS AT NASA-CSC/CDR JOHN C. STASINCEFOULS \*\*\*\*\*  
 \*\* TABLE OF PEAK AND TOTAL FLUXES SEE PERIODS: ENERGY 50-200 eV \*\*\*\*\*

DATA NUMBER	ENVELOPE	PERIOD	LONGITUDE (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	CACT TIME (HOURS)	FIELD (G)	LIE (L)	TOTAL FLUX PER ORBIT
1	4.80E 05	2/27/72	81.252	84.66	10371.44	0.14667	2.01761	5.61	1.518E 05
2	4.810E 05	2/27/72	122.076	10.54	10367.95	8.78333	0.01311	5.58	1.248E 05
3	3.507E 05	2/27/72	34.227	4.37	10367.34	14.54333	2.0152	5.68	7.058E 05
4	3.52E 05	2/27/72	172.24	5.77	10371.14	14.54333	2.01771	5.68	1.012E 05

\*\*\*\*\* ORBITA FLUX STUDY WITH AMPERSIE PARTIAL ENVIRONMENTAL VECTORS FROM 1972 TO 1979 \*\*\*\*\*  
 \*\* JNCERTAINTY FACTORS (U) APPLIED FOR THIS RUN ARE: 1.00 FOR ALL DATA, 1.00 FOR ALL DATA, 1.00 FOR ALL DATA \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES (U) AND L COMPUTED BY INAVIA OF 1972 WITH ALL DATA, 1.00 FOR ALL DATA, 1.00 FOR ALL DATA \*\*\*\*\*  
 \*\* VECTORS (U) AND L COMPUTED BY INAVIA OF 1972 WITH ALL DATA, 1.00 FOR ALL DATA, 1.00 FOR ALL DATA \*\*\*\*\*  
 \*\* FOR INFORMATION ON EXPLANATION CONTACT STASINCEFOULS AT NASA-CSC/CDR JOHN C. STASINCEFOULS \*\*\*\*\*  
 \*\* TABLE OF PEAK AND TOTAL FLUXES SEE PERIODS: ENERGY 50-200 eV \*\*\*\*\*

DATA NUMBER	ENVELOPE	PERIOD	LONGITUDE (DEG)	LATITUDE (DEG)	ALTITUDE (KM)	CACT TIME (HOURS)	FIELD (G)	LIE (L)	TOTAL FLUX PER ORBIT
1	3.158E 05	2/27/72	72.400	-22.64	10372.44	5.61333	2.00254	4.02	4.524E 10
2	3.157E 05	2/27/72	-11.2427	-23.73	10372.44	11.61000	0.02254	4.02	3.508E 10
3	3.167E 05	2/27/72	42.220	-19.14	10372.44	14.54667	0.02254	4.02	3.508E 10
4	3.174E 05	2/27/72	-34.104	-23.77	10372.44	11.61000	2.00254	4.02	3.508E 10

TABLE 30

TABLE -

NVLX115-MAX  
CIRCULAR

INCLINATION: 60 DEG  
PERIGEE: 1667 KM  
APUGEE: 1667 KM

TABLE -

NVLX115-MAX  
CIRCULAR

INCLINATION: 60 DEG  
PERIGEE: 1667 KM  
APUGEE: 1667 KM

\*\*\* EXPOSURE ANALYSIS \*\*\*

PHOTONS (E > 0.000MEV)      ELECTRONS L<sub>α</sub> (E > 0.000MEV)

PERCENT OF TOTAL LIFE-  
TIME SPENT IN FLUX-FREE  
REGIONS OF SPACE :      21.04 %      2.16 %

PERCENT OF TOTAL LIFE-  
TIME SPENT IN HIGH-  
INTENSITY REGIONS OF  
VAN ALLEN BELTS :      49.47 %      54.11 %

PERCENT OF TOTAL DAILY  
FLUX ACCUMULATED IN  
HIGH-INTENSITY REGIONS :      24.74 %      66.50 %

\*\*\*\*\*  
\* < 1 PARTICLE/CM\*\*2/SEC  
\* > 1.65 FLUX/CM\*\*2/SEC OF 1.03 PR/CM\*\*2/SEC

\* PERCENT OF TOTAL LIFETIME SPENT INSIDE AND \*

\* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT \*

INNER ZONE - I1 - \* :      67.59 %  
(1.0 < L < 2.8)

OUTER ZONE - I2 - \* :      30.35 %  
(2.8 < L < 11.0)

EXTERNAL - TE - \* :      1.67 %  
(L > 11.0)

TOTAL :      100.00 %

\* TIME IN INNER ZONE MAY BE SUBDIVIDED AS FOLLOWS:

OUTSIDE TRAPPING REGION :      0.0 %  
(1.0 < L < 1.1)

INSIDE TRAPPING REGION :      67.59 %  
(1.1 < L < 2.8)

TABLE 31

TABLE -

MAXIMUM-MAX  
CIRCULAR

INCLINATION: 60 DEG  
PERIGEE: 2553 KM  
APOGEE: 2553 KM

PERCENT OF TOTAL LIFETIME SPENT INSIDE AND \*  
\* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT \*

INTERNAL ZONE - I1- \* : 62.43 %  
(1.0 < L < 2.0)  
OUTER ZONE - I2- : 35.15 %  
(2.0 < L < 11.0)

EXTERNAL - I3- : 2.27 %  
(L > 11.0)

TOTAL : 100.00 %

\* TIME IN INNER ZONE MAY BE SUBDIVIDED AS FOLLOWS:

OUTSIDE TRAPPING REGION : 0.0 %  
(1.0 < L < 1.1)

INSIDE TRAPPING REGION : 62.43 %  
(1.1 < L < 2.0)

TABLE -

MAXIMUM-MAX  
CIRCULAR

INCLINATION: 60 DEG  
PERIGEE: 2553 KM  
APOGEE: 2553 KM

\*\*\*\*\* EXPONENT ANALYSIS \*\*\*\*\*

REGION : ELECTRONIC CC  
(1.25-0.00MEV)

PERCENT OF TOTAL LIFE-  
TIME SPENT IN HIGH-INTEN-  
SITY REGIONS \* OF

VAN ALLEN BELTS : 70.79 %

PERCENT OF TOTAL DAILY  
FLUX ACCUMULATED IN  
HIGH-INTENSITY REGIONS : 96.30 %

\* < 1 PARTICLE / CM\*\*2 / SEC

\* > 1.05 FLUX / CM\*\*2 / SEC OR 1.03 PR / CM\*\*2 / SEC

TABLE 302

TABLE -

NVLX3:S-MAX

CIRCULAR

INCLINATION: 60 DEG

PERIGEE: 3889 KM

APOGEE: 3889 KM

\*\*\*\* EXPOSURE ANALYSIS \*\*\*\*

FOTONS (E>5.000MEV) ELECTRONS IO (E>5000MEV)

PERCENT OF TOTAL LIFE-TIME SPENT IN FLUX-FREE REGIONS OF SPACE : 30.14 % 4.17 %

PERCENT OF TOTAL LIFE-TIME SPENT IN HIGH-INTENSITY REGIONS OF VAN ALLEN BELTS : 57.92 % 81.94 %

PERCENT OF TOTAL DAILY FLUX ACCUMULATED IN HIGH-INTENSITY REGIONS: 99.98 % 99.71 %

\*\*\*\*\*

\* <1 PARTICLE/CM\*\*2/SEC  
\* >1.E5 EL/CM\*\*2/SEC OR 1.E3 PD/CM\*\*2/SEC

TABLE -

NVLX3:S-MAX

CIRCULAR

INCLINATION: 60 DEG

PERIGEE: 3889 KM

APOGEE: 3889 KM

\* PERCENT OF TOTAL LIFETIME SPENT INSIDE AND \*  
\* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT \*

INNER ZONE -TI- : 54.51 %  
(1.0 < L < 2.8)

CUTER ZONE -TO- : 42.08 %  
(2.8 < L < 11.0)

EXTERNAL -TE- : 3.40 %  
(L > 11.0)

TOTAL : 100.00 %

\*TIME IN INNER ZONE MAY BE SURDIVIDED AS FOLLOWS:

OUTSIDE TRAPPING REGION : 0.0 %  
(1.0 < L < 1.1)

INSIDE TRAPPING REGION : 54.51 %  
(1.1 < L < 2.8)

TABLE 33

TABLE

TABLE

NALDAIS-NAZ  
-CIRCULAR-

INCLINATION: 60 DEG  
PERIGEE: 5180 KM  
APOGEE: 5180 KM

\* PERCENT OF TOTAL LIFETIME SPENT INSIDE AN \*  
\* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT \*

INNER ZONE -11- : 48.33 %  
11.0 < L < 3.8  
OUTER ZONE -10- : 46.67 %  
3.8 < L < 1.1

EXTERNAL -7E- : 5.00 %  
(L > 11.0)  
TOTAL : 100.00 %

\* TIME IN INNER ZONE MAY BE SUBDIVIDED AS FOLLOWS:

OUTSIDE TRAPPING REGION : 0.0 %  
(1.0 < L < 1.1)  
INSIDE TRAPPING REGION : 48.33 %  
(1.1 < L < 3.8)

\*\*\*\* EXPOSURE ANALYSIS \*\*\*\*

FCTIONS ELFCTRONS LU  
(E>5000MEV) (E>5000MEV)

PERCENT OF TOTAL LIFE-  
TIME SPENT IN FLUX-FREE  
REGIONS OF SPACE : 34.00 %

PERCENT OF TOTAL LIFE-  
TIME SPENT IN HIGH-  
INTENSITY REGIONS OF  
VAN ALLEN BELTS : 87.57 %

PERCENT OF TOTAL DAILY  
FLUX ACCUMULATED IN  
HIGH-INTENSITY REGIONS : 99.99 %

\* < 1 PARTICLE/CM\*\*2/SEC  
\* > 1000 EL/CM\*\*2/SEC OF 1.0-3 PR/CM\*\*2/SEC

TABLE 34

TABLE -

AVL 3515-MAX  
 CIRCULAR  
 INCLINATION: 60 DEG  
 RADIUS: 6369 KM  
 ALTITUDE: 6389 KM

\* PERCENT OF TOTAL LIFETIME SPENT INSIDE AND \*  
 \* OUTSIDE THE TRAPPED-PARTICLE RADIATION BELT \*

INNER BELT - I<sub>1</sub> : 42.36 %  
 (1.0 < L < 2.8)  
 OUTER BELT - I<sub>2</sub> : 51.64 %  
 (2.8 < L < 11.0)  
 EXTENSIVE - I<sub>E</sub> : 5.69 %  
 (L > 11.0)  
 TOTAL : 100.00 %

\* TIME IN WHICH TIME MAY BE SURVIVED AS FOLLOWS:

OUTSIDE TRAPPING REGION : 6.66 %  
 (1.0 < L < 1.1)  
 INSIDE TRAPPING REGION : 42.36 %  
 (1.1 < L < 2.8)

TABLE -

AVL 3515-MAX  
 CIRCULAR  
 INCLINATION: 60 DEG  
 RADIUS: 6369 KM  
 ALTITUDE: 6389 KM

\*\*\*\*\* EXPOSURE ANALYSIS \*\*\*\*\*

ELECTRONS L<sub>1</sub>  
 (0.25-0.00 MEV)

PROTONS  
 (0.25-0.00 MEV)  
 14.64 %  
 6.51 %  
 46.67 %  
 46.95 %

\*\*\*\*\*

\* < 1 PARTICLE/CM\*\*2/SEC

\* > 1000 PARTICLES/CM\*\*2/SEC OR 1.073 10\*\*10/CM\*\*2/SEC

TABLE -

TABLE -

APR 1967 - MAY 1967		APR 1967 - MAY 1967	
REGULAR		REGULAR	
INTEGRATION PERIODS		INTEGRATION PERIODS	
PERIOD	START TIME	PERIOD	START TIME
1000-1100	1000	1000-1100	1000
1100-1200	1100	1100-1200	1100
*** CUMULATIVE ANALYSIS ***			
PERCENT OF TOTAL TIME SPENT INSIDE AND OUTSIDE VAN AND SHIELDED RADIATION BELT *		ELECTRICITY CONSUMPTION (KWH)	
INSIDE VAN - YES	16.53 %		
OUTSIDE VAN - YES	72.71 %		
EXTENDED (IN 1100)	16.42 %		
TOTAL	100.00 %		
* TIME IN THESE PERIODS MAY BE SUPERVISED AS FOLLOWS:			
OUTSIDE RADIATION BELT	0.0 %		
INSIDE VAN POSITION	16.53 %		
*****			
* KI PARTICLES / CM <sup>2</sup> / SEC			
* SILENT WATCH / SEC OR 1000 HOURS / SEC			
*****			

TABLE 36



\*\*\*\*\*  
 ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS. VELLEYS APPL. ALG. - AE17 FOR SOLAR MAXIMUM \*\*\*\*\*  
 \*\*\*\*\*  
 UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APB) - UF = 1.0; FOR INNER ZONE ELECTRONS (AE6) - TIME = 1989.5 \*\*\*\*\*  
 \*\*\*\*\*  
 MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG. MODEL 4; BARRACLOUGH ET AL. I.C. - TERM 1975; TIME = 1989.5 \*\*\*\*\*  
 \*\*\*\*\*  
 MESSAGE - NOV 11-5 - MAY AN INCLINATION CODEG. AS PERIGEE - 166.2 KM. AS ARGUMENTS - 166.2 KM. AS ARGUMENTS - 166.2 KM. AS ARGUMENTS - 166.2 KM. AS ARGUMENTS \*\*\*\*\*  
 \*\*\*\*\*  
 FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINOPOULOS AT NASA-GSFC, CODE 601, GREENBELT, MARYLAND 20771. TEL. (301) 344-8067 \*\*\*\*\*  
 \*\*\*\*\*  
 \*\*\*\*\*  
 EMERGETIC SOLAR PROTON FLUENCE \*\*\*\*\*  
 \*\*\*\*\*  
 FOR CUT OFF DIPLE SHELL L5 ER \*\*\*\*\*  
 \*\*\*\*\*  
 \*\*\*\*\* (PARTICLES/CMS 2) \*\*\*\*\*  
 \*\*\*\*\*  
 \*\*\*\*\*

ENERGY LEVELS > (MEV)	*****MISSION DURATION T=60. MONTHS *****			GFOMAGNETIC SHIELDING	
	80	85	90	DIPLOFF SHELL	PERCENT TIME
10.0	5.636E 09	5.636E 09	7.515E 09	9.394E 09	1.315E 10
20.0	3.865E 09	3.865E 09	5.153E 09	6.441E 09	9.017E 09
30.0	2.850E 09	2.850E 09	3.535E 09	4.418E 09	6.495E 09
40.0	1.817E 09	1.817E 09	2.423E 09	3.028E 09	4.239E 09
50.0	1.246E 09	1.246E 09	1.661E 09	2.076E 09	2.907E 09
60.0	8.542E 08	8.542E 08	1.139E 09	1.474E 09	1.945E 09
70.0	5.857E 08	5.857E 08	7.839E 08	9.762E 08	1.327E 09
80.0	4.016E 08	4.016E 08	5.355E 08	6.583E 08	8.425E 08
90.0	2.754E 08	2.754E 08	3.671E 08	4.587E 08	6.405E 08
100.0	1.858E 08	1.858E 08	2.517E 08	3.147E 08	4.021E 08
110.0	1.252E 08	1.252E 08	1.726E 08	2.178E 08	2.871E 08
120.0	8.277E 07	8.277E 07	1.164E 08	1.513E 08	2.020E 08
130.0	6.046E 07	6.046E 07	8.115E 07	1.075E 08	1.475E 08
140.0	4.173E 07	4.173E 07	5.564E 07	7.265E 07	9.775E 07
150.0	2.822E 07	2.822E 07	3.615E 07	4.755E 07	6.477E 07
160.0	1.922E 07	1.922E 07	2.506E 07	3.273E 07	4.578E 07
170.0	1.345E 07	1.345E 07	1.744E 07	2.342E 07	3.139E 07
180.0	9.244E 06	9.244E 06	1.230E 07	1.657E 07	2.152E 07
190.0	6.352E 06	6.352E 06	8.455E 06	1.124E 07	1.476E 07
200.0	4.337E 06	4.337E 06	5.782E 06	7.822E 06	1.012E 07

TABLE 31

\*\*\*\*\*  
 \*\* ORIGINALLY STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS. VETES. APPS. AEB. LABEL FOR SOLAR MAXIMUM UNFLUX OF 1972  
 \*\* UNITS IN MV STAYS (1) APPLIED FOR THIS RUN. ARE FURTHER DETAILS. MODEL OF 1972 WITH ALL MAG. MODEL 41.0 HARBAC. OUGH ET AL. FROM 1975. TIME = 1989.5  
 \*\* MAGNETIC COORDINATES B AND L COMPUTED BY INVVARA OF 1972 WITH ALL MAG. MODEL 41.0 HARBAC. OUGH ET AL. FROM 1975. TIME = 1989.5  
 \*\* REMARKS - MIN X 2.5 - MAX AN INCLINATION. ADDRESS. 250.0 KM. BUZZER. 250.0 KM. BUZZER. 250.0 KM. BUZZER. 250.0 KM. BUZZER. 250.0 KM. BUZZER.  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT E.C. STASSI @POBOX AT NASA-GSFC, CODE 601, GREENBELT, MARYLAND 20771. TEL. (301) 344-8007  
 \*\*\*\*\*  
 \*\*\*\*\* ENERGETIC SPECTRA FROM TUNCE \*\*\*\*\*  
 \*\*\*\*\* FOR CUT OFF PARTICLES SHELL L5 \*\*\*\*\*  
 \*\*\*\*\* (PARTICLES/CM2) \*\*\*\*\*  
 \*\*\*\*\*  
 \*\*\*\*\*  
 \*\*\*\*\*  
 \*\*\*\*\*

ENERGY LEVELS (MEV)	CONFIDENCE LEVEL (%)			95	99	GEO-MAGNETIC CLIFF SHELL	SHIELDING PERCENT EXPOSURE TIME
	80	85	90				
10.0	6.651E 09	6.651E 09	8.868E 09	1.100E 10	1.552E 10	L24	19.37
20.0	4.561E 09	4.561E 09	6.081E 09	7.601E 09	1.064E 10	L24	13.19
30.0	3.122E 09	3.122E 09	4.159E 09	5.214E 09	7.294E 09	L24	9.85
40.0	2.144E 09	2.144E 09	2.859E 09	3.574E 09	5.003E 09	L27	7.22
50.0	1.470E 09	1.470E 09	1.960E 09	2.450E 09	3.430E 09		
60.0	1.008E 09	1.008E 09	1.344E 09	1.680E 09	2.350E 09		
70.0	6.912E 08	6.912E 08	9.216E 08	1.150E 09	1.615E 09		
80.0	4.730E 08	4.730E 08	6.310E 08	7.880E 08	1.104E 09		
90.0	3.250E 08	3.250E 08	4.331E 08	5.419E 08	7.580E 08		
100.0	2.220E 08	2.220E 08	2.971E 08	3.719E 08	5.190E 08		
110.0	1.520E 08	1.520E 08	2.037E 08	2.749E 08	3.790E 08		
120.0	1.040E 08	1.040E 08	1.397E 08	1.749E 08	2.440E 08		
130.0	7.180E 07	7.180E 07	9.577E 07	1.190E 08	1.670E 08		
140.0	4.920E 07	4.920E 07	6.567E 07	8.200E 07	1.140E 08		
150.0	3.377E 07	3.377E 07	4.503E 07	5.600E 07	7.870E 07		
160.0	2.315E 07	2.315E 07	3.082E 07	3.850E 07	5.400E 07		
170.0	1.580E 07	1.580E 07	2.117E 07	2.640E 07	3.700E 07		
180.0	1.080E 07	1.080E 07	1.451E 07	1.810E 07	2.540E 07		
190.0	7.400E 06	7.400E 06	9.824E 06	1.270E 07	1.770E 07		
200.0	5.110E 06	5.110E 06	6.824E 06	8.530E 06	1.194E 07		

TABLE 30

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTAL VETTES AP4, A66, A67 FOR SOLAR MAXIMUM (UNILUX) OF 1978  
 UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PHOTONS (AP4) - UF= 1.3; FOR INNER CORE ELECTRONS (UNILUX) OF 1978  
 MAGNETIC COORDINATES B AND L COMPUTED BY IMVGA OF 1972 WITH ALLMAG MODEL 4; FOR CLOUDS (AP6) - UF= 1.3; FOR CLOUDS (AP6) OF 1978  
 VEHICLE: NVLX3-S-MAX \*\* INCLINATION= 60 DEG. \*\* PERIGEE= 388.4 KM \*\* APOGEE= 384.4 KM \*\* PERIOD= 100.9 MIN \*\* TIME= 1987.5  
 FOR INFORMATION OR EXPLANATION CONTACT E.G. STASINPOULOS AT NASA-JSC, CODE 601, GREENBELT, MARYLAND 20771, TEL: 301-344-4067

ENERGY LEVELS > (MEV)	80	85	90	95	99	MAGNETIC SHIELDING PERCENT	CUTOFF SHELL	EXPOSURE TIME
10-0	0	0	0	0	0	20.13	L>4	20.13
20-0	0	0	0	0	0	10.60	L>5	10.60
30-0	0	0	0	0	0	12.44	L>6	12.44
50-0	0	0	0	0	0	15.51	L>7	15.51
70-0	0	0	0	0	0			
80-0	0	0	0	0	0			
100-0	0	0	0	0	0			
120-0	0	0	0	0	0			
140-0	0	0	0	0	0			
150-0	0	0	0	0	0			
160-0	0	0	0	0	0			
170-0	0	0	0	0	0			
180-0	0	0	0	0	0			
190-0	0	0	0	0	0			
200-0	0	0	0	0	0			

TABLE 39

\*\*\*\*\*  
 \*\* ORBITAL FLUX SLIDY WITH COMPOSITE PARTICLE ENRICHMENT: VETTS AP8; ADO; A.17 FOR SCAR MAXIMUM \*\*\*\*\*  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR POSITIONS (AP8) - UFE 1.0; FOR INFLUENCE OF STERNS (A.17) UFE 1.0 \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES B AND L COMPUTED BY INVAYA OF 1972 WITH ALL MAG. MODEL 4; BAZARAL FOR STYAL; LUTHEIN 1974; TIME 1969.5 \*\*\*\*\*  
 \*\* FOR INFORMATION OF EXPLANATION CONTACT I.G. STASSINGFULDS AT NASA-OSF/CDC/DC/011, 5000 MARYLAND, COLLEGE PARK, MD 20740 \*\*\*\*\*  
 \*\* FOR INFORMATION OF EXPLANATION CONTACT I.G. STASSINGFULDS AT NASA-OSF/CDC/DC/011, 5000 MARYLAND, COLLEGE PARK, MD 20740 \*\*\*\*\*  
 \*\*\*\*\*  
 \*\* ENEGETIC JOLAN PROTON FLUENCE \*\*\*\*\*  
 \*\* EFC CUTOFF (IPOLL) SHELL L=5 BA \*\*\*\*\*  
 \*\*\*\*\*  
 \*\* (PARTICLES/CM<sup>2</sup>) \*\*\*\*\*  
 \*\*\*\*\*

ENERGY LEVELS (TIME)	MISSISSIPPI DURATION TIME MONTHS *****					GEOMAGNETIC SHELLING		
	R0	E5	Y5	SS	TIME	TYPE	START	TIME
10000	1.0612	1.0618	1.414	1.708	10	L29	1975	11.74
15000	7.272	7.272	9.67	1.212	10	L29	1975	11.74
20000	4.587	4.587	6.69	8.311	09	L29	1975	11.74
30000	2.415	2.415	4.55	5.09	09	L29	1975	11.74
40000	1.622	1.622	3.12	3.08	09	L29	1975	11.74
50000	1.122	1.122	2.18	2.07	09	L29	1975	11.74
60000	7.122	7.122	1.00	1.00	09	L29	1975	11.74
70000	5.122	5.122	6.51	8.07	08	L29	1975	11.74
80000	2.405	2.405	4.73	5.07	08	L29	1975	11.74
90000	1.671	1.671	3.22	4.01	08	L29	1975	11.74
100000	1.151	1.151	2.27	2.78	08	L29	1975	11.74
110000	8.64	8.64	1.04	1.09	08	L29	1975	11.74
120000	5.25	5.25	7.18	8.75	07	L29	1975	11.74
130000	3.25	3.25	4.82	5.42	07	L29	1975	11.74
140000	1.73	1.73	2.27	2.20	07	L29	1975	11.74
150000	1.16	1.16	1.58	1.58	07	L29	1975	11.74
200000	5.162	5.162	1.48	1.30	06	L29	1975	11.74

**TABLE 40**

\*\*\*\*\* BLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: MILES AB8; AE6; AE17 FOR SOLAR MAXIMUM \*\*\*\*\* UNIFLX CF 1979 \*\*\*\*\*  
\*\* JNCERTAINTY FACTORS (JF) APPLIED FOR THIS RUN ARE: FCF FRCFNS (AFRT - JF= 1.0); FPD (M) ZONE ELECTRONS (AE6) UF= 170 \*\*\*\*\*  
\*\* MAGNETIC COORDINATES (U AND L) COMPUTED BY INVAVA OF 1972 WITH ALLOWA; MCDEL 4; FARRAD (DU) (U L) AL 16-YRW 1975 \* TIME= 1989.5 \*\*  
\*\*\*\*\* INFORMATION OR EXPLANATION CONTACT E. STASSINAKOPOULOS AT NASA-CSC/CID/631, CP/CD/17, MARYLAND 20771, TEL. (301) 344-8067 \*\*\*\*\*  
\*\*\*\*\* FOR INFORMATION ON PARTICLE ENVIRONMENTS (MILES AB8) CONTACT E. STASSINAKOPOULOS AT NASA-CSC/CID/631, CP/CD/17, MARYLAND 20771, TEL. (301) 344-8067 \*\*\*\*\*  
\*\*\*\*\* INFORMATION OR EXPLANATION CONTACT E. STASSINAKOPOULOS AT NASA-CSC/CID/631, CP/CD/17, MARYLAND 20771, TEL. (301) 344-8067 \*\*\*\*\*

ENERGY LEVELS > (MEV)	***** MISSION DURATION T=60 MONTHS *****			
	80	85	90	95
10.0	1.288E 10	1.288E 10	1.718E 10	2.147E 10
20.0	8.833E 09	8.833E 09	1.178E 10	1.472E 10
30.0	6.057E 09	6.057E 09	8.076E 09	1.009E 10
40.0	4.153E 09	4.153E 09	5.537E 09	6.921E 09
50.0	2.847E 09	2.847E 09	3.797E 09	4.746E 09
60.0	1.952E 09	1.952E 09	2.603E 09	3.254E 09
70.0	1.335E 09	1.335E 09	1.785E 09	2.231E 09
80.0	9.179E 08	9.179E 08	1.224E 09	1.620E 09
90.0	6.294E 08	6.294E 08	8.392E 08	1.049E 09
100.0	4.316E 08	4.316E 08	5.754E 08	7.193E 08
110.0	2.959E 08	2.959E 08	3.945E 08	4.932E 08
120.0	2.029E 08	2.029E 08	2.705E 08	3.382E 08
130.0	1.351E 08	1.351E 08	1.855E 08	2.319E 08
140.0	9.535E 07	9.535E 07	1.272E 08	1.590E 08
150.0	6.541E 07	6.541E 07	9.721E 07	1.090E 08
160.0	4.466E 07	4.466E 07	6.400E 07	7.474E 07
170.0	3.074E 07	3.074E 07	4.100E 07	5.125E 07
180.0	2.106E 07	2.106E 07	2.811E 07	3.514E 07
190.0	1.449E 07	1.449E 07	1.928E 07	2.409E 07
200.0	9.913E 06	9.913E 06	1.322E 07	1.652E 07

GEOMAGNETIC SHELL	CUT OFF	SHIELDING PERCENT EXPOSURE TIME
L29	L29	37.71
L36	L36	25.56
L37	L37	18.80
L37	L37	13.82

TABLE 41



ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTAL VELOCITIES APPLIED TO PROTONS (APB) - JFE 1.01 FOR INNER ZONE FLECTIONS (AE6) - OF = 1.0  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: 1972 WITH ALL MAG, MODEL 4; SARRACLOUGH ET AL, 1971-IPM 1975 & TIME = 1989.5  
 \*\* MAGNETIC COORDINATES 9 AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG, MODEL 4; SARRACLOUGH ET AL, 1971-IPM 1975 & TIME = 1989.5  
 \*\* MAGNETIC COORDINATES 9 AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG, MODEL 4; SARRACLOUGH ET AL, 1971-IPM 1975 & TIME = 1989.5  
 \*\* FOR INFORMATION IN EXPLANATION CONTACT E.C. STASSINPOULOS AT NASA-GFC/CDUF 001; GREENBELL, MARYLANO 20771; TEL: (301)-348-8067 \*\*

\*\*\* DCS AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\*\*  
 \*\*\* MISSION DURATION: 5,000 YEARS \*\*\*

SHIELD THICKNESS (ALUMINUM) (G/MCM <sup>2</sup> )	ELECTRONS*		URLEMSSIM- AFLJG		TOTAL IONS		TOTAL DOSE			
	T (MILS)	INNER ZN: (RAD-S-AL)	OUTER ZN: (RAD-S-AL)	TOTAL (RAD-S-AL)	TRAPPED** (RAD-S-AL)	SOLAR+ (RAD-S-AL)	TOTAL (RAD-S-AL)	ALL SOURCES (RAD-S-AL)		
0.01	0.04	1.0	2.257E 07	6.399E 05	2.321E 07	4.854E 03	6.600E 05	9.988E 02	6.57E 05	2.392E 07
0.03	0.07	3.0	7.78E 06	2.882E 05	1.072E 07	1.677E 03	2.750E 05	1.035E 03	2.54E 05	8.33E 07
0.05	0.11	4.0	5.08E 06	2.587E 05	8.072E 06	1.571E 03	2.605E 05	1.035E 03	2.60E 05	8.33E 06
0.07	0.15	5.0	3.43E 06	1.854E 05	5.205E 06	1.071E 03	1.865E 05	1.035E 03	1.87E 05	5.81E 06
0.09	0.19	6.0	2.38E 06	1.244E 05	3.539E 06	7.29E 02	1.205E 05	1.035E 03	1.24E 05	2.71E 06
0.09	0.22	9.0	1.69E 06	1.340E 05	2.826E 06	3.03E 02	1.035E 03	1.035E 03	1.50E 05	1.08E 06
0.09	0.24	10.0	1.22E 06	1.16E 05	2.345E 06	2.81E 02	1.350E 05	1.035E 03	1.37E 05	1.48E 06
0.09	0.27	13.0	9.09E 05	1.037E 05	1.913E 06	2.89E 02	1.205E 05	1.035E 03	1.24E 05	1.48E 06
0.10	0.33	15.0	6.87E 05	9.31E 04	1.408E 06	2.24E 02	1.120E 05	1.035E 03	1.17E 05	9.70E 05
0.20	0.74	25.0	1.06E 05	1.532E 04	1.517E 05	1.58E 02	6.90E 04	8.76E 02	6.55E 04	2.27E 05
0.30	1.41	40.0	3.98E 04	4.66E 03	4.66E 04	8.30E 01	5.30E 04	7.06E 02	5.12E 04	1.88E 05
0.40	1.88	58.0	2.04E 04	1.99E 03	2.719E 04	5.07E 01	3.20E 04	5.82E 02	3.12E 04	7.90E 04
0.50	2.48	73.0	1.20E 04	1.170E 03	2.281E 04	5.88E 01	2.02E 04	5.02E 02	3.57E 04	5.84E 04
0.60	3.22	87.0	7.51E 03	7.192E 02	1.420E 04	4.30E 01	1.40E 04	3.46E 02	3.12E 04	4.61E 04
0.80	5.66	117.0	3.12E 03	3.53E 02	6.657E 03	3.42E 01	1.03E 04	3.46E 02	2.50E 04	3.27E 04
1.00	8.70	146.0	1.24E 03	1.907E 02	3.148E 03	2.25E 01	2.35E 04	2.83E 02	2.26E 04	2.60E 04
1.20	12.0	182.0	5.02E 02	8.84E 01	1.197E 03	2.06E 01	1.90E 04	2.92E 02	2.64E 04	2.14E 04
1.50	2.56	216.0	5.06E 01	3.85E 01	4.358E 02	1.53E 01	1.75E 04	1.85E 02	1.81E 04	1.87E 04
1.75	6.48	255.0	6.35E 01	1.47E 01	1.537E 02	1.81E 01	1.64E 04	1.56E 02	1.67E 04	1.70E 04
2.00	9.41	292.0	6.67E 01	4.07E 01	4.07E 01	1.50E 01	1.50E 04	1.50E 02	1.54E 04	1.58E 04
2.50	9.26	365.0	1.67E 01	3.08E 00	3.08E 00	1.24E 01	1.30E 04	9.67E 01	1.34E 04	1.37E 04
3.00	11.11	432.0	0.0	5.03E 00	5.03E 00	1.64E 01	1.20E 04	1.64E 01	1.21E 04	1.22E 04
3.50	12.96	510.0	0.0	1.291E 00	1.291E 00	9.45E 01	1.09E 04	5.79E 01	1.10E 04	1.11E 04
4.00	14.81	583.0	0.0	0.0	0.0	4.33E 01	1.08E 04	4.65E 01	1.01E 04	1.02E 04
4.50	16.67	656.0	0.0	0.0	0.0	7.37E 01	9.39E 03	3.79E 01	9.47E 03	9.47E 03
5.00	18.52	729.0	0.0	0.0	0.0	6.83E 01	8.63E 03	3.09E 01	8.64E 03	8.73E 03
5.50	20.37	802.0	0.0	0.0	0.0	5.25E 01	7.52E 03	2.15E 01	7.54E 03	7.60E 03
6.00	22.22	875.0	0.0	0.0	0.0	4.26E 01	5.91E 03	1.18E 01	5.92E 03	5.96E 03
6.50	24.07	948.0	0.0	0.0	0.0	3.27E 01	4.26E 03	4.80E 00	4.81E 03	4.84E 03
10.00	37.04	1458.0	0.0	0.0	0.0	3.287E 01	3.287E 03	3.266E 00	4.81E 03	4.84E 03

\* ELECTRON MODELS:  
 AE6: INNER ZONE-SCLAP MAX  
 NO UNCERTAINTY FACILIA WAS APPLIED TO THE MODEL DATA.  
 SOLPRO: SOLAR FLARE PROTONS AT 1 AU  
 (UNATTENUATED, INTERPLANETARY)  
 FOR CUTOFF DIPOLE SHELL OF 5 E.A.R.  
 \*\* PROTON MODEL:  
 FOR RATE COEFFICIENTS OF ALL EVENTS =  
 84.82X GEOMAGNETIC SHIELDING APPLIED

AF17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.  
 FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER &  
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN  
 EXISTING DATA SETS. THE AE17-HI FAVORS VAMPOLA'S FIT  
 TO OLV-19 DATA WHILE AE17-LO IS MORE REPRESENTATIVE OF  
 ALL THE DATA SETS PRESENTLY AVAILABLE TO ASSDC.  
 \*\* THE AE17-LO VERSION WAS USED FOR THESE CALCULATIONS \*\*

\*\* PROTON MODEL:  
 APB-MAG: TRAPPED PROTONS-SCLAP MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR  
 PROTON SPECTRA EITHER TOWARDS LOWER NOR  
 TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS  
 USED IN THE CONSTRUCTION OF THE MODEL IS A  
 LIMITED MEASUREMENTS MADE DURING THE 20TH SOLAR  
 CYCLE: 1964-1975) DO NOT CONTAIN INFORMATION  
 FOR ESTO AND C200 MEV

TABLE 43

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VETTES APPL AEG, AE17 FOR SOLAR MAXIMUM UNIFLUX OF 1979  
 \*\* UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FLR PROT CNS (AP5) - JF= 1.3; FUP INVR ZONE FLCTRNS (AE6) - UF= 1.0  
 \*\* MAGNETIC COORDINATES B AND L COMPUTED BY INVAFA OF 1972 WITH ALL MAG. MODEL 4: HARRACLUCH ET AL. 16-TRM 1975 \* TIME = 1989.5  
 \*\* VEHICLE - NAVY'S MAX. AN INCLINATION: 60 DEG. 350 KM. ADJUGE: 250 KM. AB. BL. DUBLI. TAPE: JDSAA6 AA. RE. COL. 2-349  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINOPoulos AT NASA-GSFC, CODE 601, GREENBELT, MARYLAND 20771, TEL. (301)-344-8007

\*\*\* DCS AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\*\*  
 \*\*\* MESSAGE DURATION: 5.0000 YEAR(S) \*\*\*

SHIELD THICKNESS (ALUMINUM) S (CM/CM2)	T (MILLS)	ELECTRONS*			PROTONS			TOTAL DOSE	
		INNER ZN. (RAUS-AL)	OUTER ZN. (RAUS-AL)	TOTAL (RAUS-AL)	TRAPPED** (RAUS-AL)	SOLAR** (RAUS-AL)	TOTAL (RAUS-AL)	ALL SOURCES (RAUS-AL)	
0.01	0.04	7.905E 07	1.199E 06	9.025E 07	2.414E 04	4.844E 06	1.175E 03	4.845E 06	8.513E 07
0.02	0.07	4.384E 07	7.425E 05	4.458E 07	2.693E 04	2.798E 06	1.197E 03	2.798E 06	4.781E 07
0.03	0.11	2.717E 07	5.364E 05	2.770E 07	2.162E 04	2.150E 06	1.210E 03	2.152E 06	2.988E 07
0.04	0.15	1.785E 07	4.174E 05	1.825E 07	1.783E 04	1.765E 06	1.219E 03	1.766E 06	2.001E 07
0.05	0.19	1.210E 07	3.430E 05	1.248E 07	1.494E 04	1.512E 06	1.225E 03	1.513E 06	1.396E 07
0.06	0.22	8.450E 06	2.859E 05	8.736E 06	1.274E 04	1.323E 06	1.228E 03	1.324E 06	1.007E 07
0.07	0.26	6.045E 06	2.460E 05	6.204E 06	1.103E 04	1.167E 06	1.230E 03	1.168E 06	7.668E 06
0.08	0.30	4.419E 06	2.156E 05	4.635E 06	9.407E 03	1.041E 06	1.228E 03	1.042E 06	5.690E 06
0.09	0.33	3.295E 06	1.916E 05	3.490E 06	8.004E 03	9.343E 05	1.225E 03	9.343E 05	4.435E 06
0.10	0.37	2.514E 06	1.724E 05	2.685E 06	7.474E 03	8.498E 05	1.217E 03	8.471E 05	3.581E 06
0.20	0.74	4.131E 05	8.368E 04	4.967E 05	3.962E 03	4.045E 05	1.034E 02	4.045E 05	9.062E 05
0.30	1.11	1.601E 05	4.565E 04	2.095E 05	2.732E 03	2.917E 05	6.265E 02	2.915E 05	4.640E 05
0.40	1.48	8.446E 04	3.093E 04	1.154E 05	2.105E 03	1.764E 05	5.916E 02	1.771E 05	2.946E 05
0.50	1.85	5.069E 04	1.993E 04	7.061E 04	1.724E 03	1.764E 05	5.920E 02	1.761E 05	2.085E 05
0.60	2.22	3.208E 04	1.328E 04	4.536E 04	1.466E 03	1.411E 05	4.074E 02	1.411E 05	1.584E 05
0.80	2.96	1.743E 04	8.519E 03	1.995E 04	1.136E 03	8.242E 04	4.054E 02	8.242E 04	1.039E 05
1.00	3.70	1.46E 04	5.516E 03	8.770E 03	9.233E 02	6.742E 04	3.343E 02	6.775E 04	7.745E 04
1.25	4.03	1.214E 04	4.640E 03	7.605E 03	7.805E 02	5.942E 04	2.677E 02	5.630E 04	6.032E 04
1.50	5.56	1.184E 04	7.066E 02	8.950E 02	2.497E 02	4.921E 04	2.193E 02	4.93E 04	5.097E 04
1.75	6.48	2.229E 03	2.687E 02	2.910E 02	5.673E 02	4.088E 04	1.836E 02	4.427E 04	4.513E 04
2.00	7.41	2.285E 03	8.844E 01	9.075E 01	5.950E 02	3.997E 04	1.542E 02	4.305E 04	4.088E 04
2.50	9.26	6.396E 03	5.362E 00	5.369E 00	4.165E 02	3.387E 04	1.142E 02	3.358E 04	3.443E 04
3.00	11.11	4.32E 03	1.022E 01	1.022E 01	3.562E 02	2.563E 04	8.82E 01	2.532E 04	3.008E 04
3.50	12.96	510. 00	0.676E-05	6.676E-05	3.119E 02	2.655E 04	6.841E 01	2.652E 04	2.693E 04
4.00	14.81	582. 00	0.00	0.00	2.775E 02	2.415E 04	5.506E 01	2.41E 04	2.449E 04
4.50	16.67	656. 00	0.00	0.00	2.499E 02	2.215E 04	4.467E 01	2.21E 04	2.242E 04
5.00	18.52	729. 00	0.00	0.00	2.268E 02	2.039E 04	3.654E 01	2.03E 04	2.065E 04
6.00	24.22	875. 00	0.00	0.00	1.950E 02	1.752E 04	2.541E 01	1.75E 04	1.775E 04
8.00	29.63	1167. 00	0.00	0.00	1.430E 02	1.342E 04	1.319E 01	1.34E 04	1.361E 04
10.00	37.04	1458. 00	0.00	0.00	1.080E 02	1.072E 04	7.395E 00	1.074E 04	1.084E 04

\* ELECTRON MODELS:  
 AE6: INNER ZONE-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 \* SOLAR PROTON MODEL:  
 SOLPRO: SOLAR FLARE PROTONS AT 1 AU  
 UNATTENUATED, INTERPLANETARY  
 FOR CUTOFF DIPOLE SHELL OF 5 F.P.  
 FOR TRU-60-MD-10-9021: \* OF AL EVENTS=4  
 PERCENT GEOMAGNETIC SHIELDING APPLIED

AE17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.  
 FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS JPEL 6  
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN  
 EXISTING DATA SETS. THE AE17-HI FAVORS VAMPALA'S FIT  
 TO ALL DATA WHILE AE17-LO IS MORE RESEMBLANTIAL TO  
 ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSDC.  
 \*\* THE AE17-LO VERSION WAS USED FOR THESE CALCULATIONS \*\*

\*\* PROTON MODEL:  
 AP8-MAC: TRAPPED PROTONS-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR  
 PROTON SPECTRA EITHER TOWARDS LOWER OR  
 TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS  
 USED IN THE CONSTRUCTION OF THE MODEL IS ATEL-  
 LITE MEASUREMENTS MADE DURING THE 20TH SOLAR  
 CYCLE (1964-1975) DO NOT CONTAIN INFORMATION  
 FOR ENERgies ABOVE 1.5 MEV.

TABLE 44





ORBITAL FLUX STUDY WITH COMPLETE PARTICLE ENVIRONMENTS: VETTER APRI AGO, A17 FOR SOLAR MAXIMUM (LATE 1961) LATE 1961  
 UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APR) - UF 1.0; FOR IONS (APR) - UF 1.0; FOR ELECTRONS (APR) - UF 1.0  
 MAGNETIC COORDINATES B AND L COMPUTED BY RUN ARE: 1572 WITH ALLMAY, MODEL 4; BARCAGLIUSSO ET AL. 168-TRM 1973; JIMEZ 1589-5  
 MAGNETIC COORDINATES B AND L COMPUTED BY RUN ARE: 1572 WITH ALLMAY, MODEL 4; BARCAGLIUSSO ET AL. 168-TRM 1973; JIMEZ 1589-5  
 FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSINOFF AT NASA-GSFC, CLD/C 311, GREENBELT, MARYLAND 20771-1131-344-3467

\*\*\* CGSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SP10C \*\*\*  
 \*\*\* MISSION DURATION: 5.0000 YEAR(S) \*\*\*

SHIELD THICKNESS (ALUMINUM)	T (GCMAGNETIC UNIT)	INNER ZONE (RAD-S-ALL)	ELECTRONS OUTER ZONE (RAD-S-ALL)	TOTAL (RAD-S-ALL)	GRMSSTR- ALUMINUM TOTAL (RAD-S-ALL)	TRAPFAC (RAD-S-ALL)	SOLAR (RAD-S-ALL)	PROTONS (RAD-S-ALL)	TOTAL (RAD-S-ALL)	TOTAL D-CURVES (RAD-S-ALL)
0.01	0.04	1.441E 08	2.555E 06	1.477E 09	6.321E 04	6.321E 04	1.600E 07	6.321E 04	6.321E 04	2.170E 08
0.02	0.07	7.257E 07	2.251E 06	9.082E 07	6.939E 04	2.928E 07	1.600E 07	6.939E 04	2.928E 07	7.170E 07
0.03	0.11	4.245E 07	1.635E 06	5.009E 07	3.952E 04	2.813E 07	1.600E 07	3.952E 04	2.813E 07	4.628E 07
0.04	0.15	3.180E 07	1.284E 06	3.509E 07	3.235E 04	1.742E 07	1.600E 07	3.235E 04	1.742E 07	3.424E 07
0.05	0.19	2.171E 07	1.022E 06	2.275E 07	2.735E 04	1.142E 07	1.600E 07	2.735E 04	1.142E 07	2.507E 07
0.06	0.22	1.524E 07	8.678E 05	1.613E 07	2.334E 04	8.025E 06	1.600E 07	2.334E 04	8.025E 06	1.807E 07
0.07	0.26	1.098E 07	7.265E 05	1.168E 07	2.021E 04	5.627E 06	1.600E 07	2.021E 04	5.627E 06	1.446E 07
0.08	0.30	7.548E 06	6.233E 05	8.619E 06	1.771E 04	4.027E 06	1.600E 07	1.771E 04	4.027E 06	1.146E 07
0.09	0.33	5.255E 06	5.356E 05	6.457E 06	1.571E 04	3.024E 06	1.600E 07	1.571E 04	3.024E 06	8.764E 06
0.10	0.37	3.828E 06	4.600E 05	4.908E 06	1.409E 04	2.341E 06	1.600E 07	1.409E 04	2.341E 06	6.664E 06
0.20	0.74	2.425E 05	2.520E 05	3.045E 05	8.705E 03	4.011E 05	1.600E 07	8.705E 03	4.011E 05	3.393E 05
0.30	1.11	1.241E 04	1.244E 05	1.108E 05	4.712E 03	2.141E 05	1.600E 07	4.712E 03	2.141E 05	2.326E 05
0.40	1.46	1.265E 04	5.205E 04	1.139E 05	2.970E 03	1.110E 05	1.600E 07	2.970E 03	1.110E 05	2.051E 05
0.50	1.82	6.267E 03	3.065E 04	6.922E 04	2.835E 03	1.110E 05	1.600E 07	2.835E 03	1.110E 05	1.825E 05
0.60	2.17	4.443E 03	1.877E 04	4.443E 04	2.420E 03	6.474E 04	1.600E 07	2.420E 03	6.474E 04	1.625E 05
0.80	3.52	1.258E 03	1.277E 04	2.027E 04	1.420E 03	4.320E 04	1.600E 07	1.420E 03	4.320E 04	1.170E 05
1.00	5.00	3.840E 02	1.028E 04	4.068E 04	1.157E 03	3.237E 04	1.600E 07	1.157E 03	3.237E 04	8.512E 04
1.25	6.87	2.045E 02	8.261E 03	2.561E 04	8.261E 03	2.561E 04	1.600E 07	8.261E 03	2.561E 04	6.014E 04
1.50	9.03	1.425E 02	6.901E 03	2.001E 04	6.901E 03	1.922E 04	1.600E 07	6.901E 03	1.922E 04	4.585E 04
1.75	11.42	9.230E 01	5.801E 03	2.259E 04	4.820E 03	1.075E 04	1.600E 07	4.820E 03	1.075E 04	3.510E 04
2.00	14.01	5.730E 01	4.871E 03	2.259E 04	3.490E 03	7.695E 03	1.600E 07	3.490E 03	7.695E 03	2.510E 04
2.50	17.26	3.175E 01	3.084E 03	1.510E 04	2.851E 03	6.890E 03	1.600E 07	2.851E 03	6.890E 03	1.825E 04
3.00	21.11	1.811E 01	2.022E 03	9.582E 03	2.420E 03	4.931E 03	1.600E 07	2.420E 03	4.931E 03	1.350E 04
4.00	28.81	8.621E 00	1.300E 03	6.000E 03	1.600E 03	3.200E 03	1.600E 07	1.600E 03	3.200E 03	8.621E 03
5.00	38.12	4.500E 00	8.000E 02	4.000E 03	1.000E 03	2.000E 03	1.600E 07	1.000E 03	2.000E 03	4.500E 03
6.00	49.43	2.300E 00	5.000E 02	2.500E 03	6.000E 02	1.500E 03	1.600E 07	6.000E 02	1.500E 03	2.300E 03
8.00	66.64	1.100E 00	3.000E 02	1.500E 03	3.000E 02	9.000E 02	1.600E 07	3.000E 02	9.000E 02	1.100E 03
10.00	88.85	5.000E 00	1.500E 02	7.500E 02	1.500E 02	4.500E 02	1.600E 07	1.500E 02	4.500E 02	5.000E 03

\* ELECTRON MODELS:  
 AEG: INNER ZONE-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA

\* SOLAR PROTON MODEL:  
 SUBJECT: SOLAR FLARE PROTONS AT 1 AU  
 (UNIDENTIFIED, INTERPLANETARY)  
 FOR CUTOFF RADIUS SHELL OF 5 R.E.  
 FOR TALKING POINTS: \* OF ALL EVENTS \*\*  
 76.503 COLUMBIANIC SH LIVING APLETT

NOTE: Q DENOTES THE DEGREE OF CONFIDENCE ENT WIGHTS  
 TO ASSIGN TO RESULTS, NAMED THAT FOR THE  
 SPECIFIC MISSION DURATION THE CALCULATION  
 FLUNCES ARE THE SMALLEST VALUES WHICH  
 NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED  
 INTENSITIES

IT IS NOT ADVISABLE TO EXTRAPOLATE TO SCALE  
 PROTON SPECTRA EITHER TOWARDS LOWER AGS  
 TOWARDS HIGHER ENERGIES BECAUSE OF DATA SETS  
 USED IN THE DETERMINATION OF THE MODEL (SABILE  
 LIFE MEASUREMENTS MADE DURING THE 2000 SEC  
 CYCLE: 1968-1973) OR TO OBTAIN INFORMATION  
 FOR P-10 AND P-2000 M.V.

\*\* PROTON MODEL:  
 APP-MAG: TRAPPED PROTONS-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA



\*\*\*\*\* FLUX STUDY WITH COMPLETE PARTICLE INTERCOMPARISON \*\*\*\*\* UNIFLUX OF 1979  
 \*\* JACENTIVITY FACTORS (F) APPLIED TO THIS ANALYSIS. UNIFLUX OF 1979. ELECTRONICS (AE6) - UFE 1.0  
 \*\* MAGNETIC COEFFICIENTS (M) APPLIED TO THIS ANALYSIS. UNIFLUX OF 1979. ELECTRONICS (AE6) - UFE 1.0  
 \*\* FOR INFORMATION ON CALIBRATION CONTACT THE STATION CONTACT AT NASA-CRDC (301) 344-8667  
 \*\*\*\*\*

SHELL ID	RADIUS (CM)	THICKNESS (CM)	L-CIRCUIT		FREMONT		TOTAL		TOTAL DOSE (RAD5-AL)	ALL SOURCES (RAD5-AL)
			AREA (CM2)	PERCENT	AREA (CM2)	PERCENT	AREA (CM2)	PERCENT		
0.01	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.02	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.03	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.04	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.05	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.06	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.07	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.08	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.09	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.10	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.11	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.12	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.13	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.14	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.15	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.16	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.17	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.18	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.19	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.20	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.21	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.22	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.23	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.24	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.25	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.26	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.27	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.28	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.29	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.30	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.31	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.32	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.33	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.34	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.35	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.36	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.37	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.38	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.39	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.40	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.41	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.42	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.43	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.44	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.45	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.46	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.47	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.48	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.49	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00
0.50	0.10	1.00	1.00	0.00	1.00	0.00	1.00	1.00	1.00	1.00

\*\*\*\*\* ELECTRON MODEL \*\*\*\*\*  
 A60: INVER ZIMP-1.047 MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 A61: OUTER ZONE-INTER-IM MODEL WITHOUT SOLAR CYCLE CORRECTION.  
 FOR ENERGIES ABOVE 1.0 MEV, THIS MODEL CONTAINS UPPER &  
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN  
 EXISTING DATA SETS. THE A617-MI FAVORS VANFLEET'S FIT  
 TO THE DATA SETS AVAILABLE TO NSR.  
 ALL THE DATA SETS ARE SENSITIVELY AVAILABLE TO NSR.  
 \*\* THE REPORT VERSION WAS USED FOR THESE CALCULATIONS \*\*  
 \*\* ELECTRON MODEL:  
 ADD-MAGN-LEADEN-ALUMINUM-SILICON-MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 \*\*\*\*\*

TABLE 48

ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: VETLS APRI AED. AE17 FLX SOLAR MAXIMUM UNIFLUX OF 1979  
 UNCERTAINTY FACTORS (UF) APPLIED FOR THIS RUN ARE: FOR PROTONS (APR) - UF = 1.0; FOR INNER ZONE ELECTRONS (AF6) - UF = 1.0  
 MAGNETIC COORDINATES B AND L COMPUTED BY INVARA OF 1972 WITH ALL MAG. MODEL 4: BAKKAC. JUGH ET AL. 16-TRM 1975 \* TIME = 1989.5  
 REMIC. - MAXIMUM INCLINATION CODES AS DESIGNATED BY AADGES. 1667M - BAL. CABLI TAPE - J05A46 - BAL. DEVI00 - 1-985  
 FOR INFORMATION OR EXPLANATION CONTACT E.G. STASSTADPOULOS AT NASA-SCF, CODE 601, GREENBELT, MARYLAND 20771. TEL. (301)-344-8067

DOSE IN SEMI-INFINITE ALUMINUM MEDIA  
 \*\*\* MISSION DURATION: 5.0000 YEAR(S) \*\*\*

SHIELD THICKNESS (ALUMINUM)	T	INNER ZONE (RAD/S-ALL)	OUTER ZONE (RAD/S-ALL)	ELECTRONS (RAD/S-ALL)	BREMSTR. TOTAL (RAD/S-ALL)	TRAPPED** (RAD/S-ALL)	PROTONS		TOTAL DOSE (RAD/S-ALL)
							SOLAR+ (RAD/S-ALL)	TOTAL (RAD/S-ALL)	
0.01	0.04	1.	3.468E 07	9.742E 05	3.566E 37	6.760E 05	9.988E C2	6.970E 05	3.637E 07
0.02	0.07	3.	1.924E 07	6.002E 05	1.294E 37	3.980E 05	1.018E C2	3.980E 05	2.021E 07
0.03	0.11	4.	1.188E 07	4.320E 05	1.232E 37	2.055E 05	1.025E C2	2.055E 05	1.259E 07
0.04	0.13	5.	7.743E 06	3.351E 05	8.091E 36	2.163E 05	1.033E C2	2.174E 05	8.335E 06
0.05	0.15	7.	5.725E 06	2.722E 05	5.897E 36	1.864E 05	1.038E C2	1.875E 05	5.690E 06
0.06	0.22	9.	3.619E 06	2.282E 05	4.947E 36	1.655E 05	1.041E C2	1.666E 05	4.018E 06
0.08	0.28	10.	2.884E 06	1.850E 05	4.221E 36	1.490E 05	1.043E C2	1.500E 05	2.914E 06
0.08	0.33	12.	1.956E 06	1.712E 05	2.327E 36	1.355E 05	1.041E C2	1.370E 05	2.169E 06
0.09	0.37	13.	1.371E 06	1.518E 05	1.523E 36	1.250E 05	1.038E C2	1.261E 05	1.652E 06
0.10	0.41	15.	1.034E 06	1.363E 05	1.170E 36	1.160E 05	1.034E C2	1.170E 05	1.290E 06
0.20	0.74	29.	1.364E 05	6.526E 04	2.216E 35	6.901E 04	9.746E C2	6.564E 04	2.928E 05
0.30	1.11	44.	5.309E 04	3.945E 04	9.512E 34	3.537E 04	7.003E C2	5.127E 04	1.483E 05
0.50	1.48	78.	2.937E 04	2.384E 04	5.321E 34	4.062E 04	5.869E C2	4.120E 04	9.512E 04
0.75	1.85	75.	1.720E 04	1.529E 04	3.249E 34	3.445E 04	5.825E C2	3.518E 04	6.815E 04
1.00	2.22	87.	1.022E 04	1.015E 04	2.082E 34	3.061E 04	4.334E C2	3.115E 04	5.242E 04
1.50	3.06	117.	4.758E 03	4.954E 03	9.409E 33	2.924E 04	3.465E C2	2.956E 04	3.563E 04
1.99	3.70	146.	1.775E 03	2.667E 03	4.841E 33	2.235E 04	2.833E C2	2.264E 04	2.743E 04
2.50	5.09	192.	7.757E 02	1.253E 03	1.684E 33	1.980E 04	2.498E C2	2.498E 04	2.198E 04
1.75	5.36	242.	7.257E 01	5.409E 02	6.135E 32	1.795E 04	1.859E C2	1.813E 04	1.998E 04
3.00	7.40	285.	9.187E 00	2.075E 02	2.106E 32	1.654E 04	1.556E C2	1.670E 04	1.712E 04
2.50	9.21	285.	9.688E -01	6.948E 01	7.040E 31	1.535E 04	1.306E C2	1.548E 04	1.574E 04
3.00	9.25	365.	2.621E -03	9.489E 00	4.492E 31	1.353E 04	9.677E C1	1.376E 04	1.375E 04
3.00	11.11	510.	0.0	7.624E -02	7.624E -02	1.205E 04	7.450E C1	1.216E 04	1.230E 04
2.00	12.90	619.	0.0	1.985E -05	1.985E -05	1.099E 04	5.757E C1	1.114E 04	1.116E 04
4.00	14.81	953.	0.0	3.0	3.0	1.009E 04	4.665E C1	1.013E 04	1.024E 04
5.00	18.57	950.	0.0	0.0	0.0	9.309E -03	3.745E C1	9.347E 03	9.442E 03
2.00	22.32	852.	0.0	0.0	0.0	8.636E 03	3.086E C1	8.667E 03	8.754E 03
3.00	26.63	1147.	0.0	0.0	0.0	7.527E 03	2.153E C1	7.545E 03	7.619E 03
10.00	37.04	1458.	0.0	0.0	0.0	5.911E 03	1.114E C1	5.922E 03	5.978E 03
						4.804E 03	6.266E 00	4.810E 03	4.853E 03

ELECTRON MODEL 3:  
 A66: INNER ZONE-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 SOLPRC: SOLAR FLARE PHOTONS AT 1 AU  
 UNATTENUATED (DIRECTIONAL) INCREASINGLY  
 FOR CUTOFF DIPPLE SHELL OF 5 E.R.

AE17: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.  
 FOR ENERGIES ABOVE 1.5 MEV, THIS MODEL CONTAINS UPPER  
 LOWER ENERGY VALUES TO ACCOUNT FOR OTSCHEP-KOENIG ET AL.  
 EXISTING DATA SETS. THE AE17-11 FAVORS VAMBERG'S  
 10-DATA-19 DATA WHILE AE17-10 FAVORS BRENKENTHAL'S  
 ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSDC.  
 \*\* THE AE17-10 VERSION WAS USED FOR THESE CALCULATIONS \*\*

PROTON MODEL:  
 A66-MAX: TRAPPED PROTONS-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR  
 PROTON SPECTRA EITHER TOWARDS LOWER ACP  
 TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS  
 USED IN THE CONSTRUCTION OF THE MODEL IS A  
 CYCLE 1564-1572) DO NOT CONTAIN INFORMATION  
 FOR ENERGY AND FLUX

\*\*\*\*\*  
 \*\* ORIGIN: MAXIMUM WITH COMPOSITE PARTICLE ENVIRONMENTS; VELOCITY: ABE17 FOR SOLAR MAXIMUM \*\*\*\*\*  
 \*\* UNCELESTRIAL FORTS AND APPLIED FOR 1975 WITH ALL WAGON MODELS \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES AND COMPUTED BY INVAR OF 1975 WITH ALL WAGON MODELS \*\*\*\*\*  
 \*\* MAGNETIC COORDINATES AND COMPUTED BY INVAR OF 1975 WITH ALL WAGON MODELS \*\*\*\*\*  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT E. S. STASSI@POLCALS AT NASA-JSCFC CODE 601 GREENBAY, MARILAND 20771 TEL: (301) 344-8067 \*\*\*\*\*

\*\*\*\*\*  
 \*\* DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \*\*\*\*\*  
 \*\* MISSION DURATION: 5.0000 YEARS \*\*\*\*\*

SHIELD THICKNESS S (ALUMINUM)	ELECTRONS*				BREMSTR- RADIATION		PROTONS		TOTAL DOSE	
	LEM/CM/HR/23	INNER ZONE (RADS-AU)	CUTTER ZONE (RADS-AU)	TOTAL ZONE (RADS-AU)	TRAPPED** (RADS-AU)	SOLAR** (RADS-AU)	SOLAR** (RADS-AU)	TOTAL (RADS-AU)	TOTAL (RADS-AU)	ALL SOURCES (RADS-AU)
0.01	0.04	1.215E 08	1.826E 06	1.233E 08	5.692E 04	1.175E 03	4.944E 06	1.042E 06	4.944E 06	1.282E 08
0.02	0.07	8.710E 07	1.121E 06	8.622E 07	3.996E 04	1.197E 03	2.798E 06	1.225E 03	2.798E 06	7.106E 07
0.03	0.11	4.147E 07	8.044E 05	4.227E 07	2.996E 04	1.210E 03	2.150E 06	1.220E 03	2.150E 06	4.455E 07
0.04	0.15	2.711E 07	6.225E 05	2.774E 07	2.996E 04	1.219E 03	1.765E 06	1.219E 03	1.765E 06	2.953E 07
0.05	0.19	1.839E 07	5.048E 05	1.889E 07	1.916E 04	1.212E 03	1.512E 06	1.212E 03	1.512E 06	2.042E 07
0.06	0.22	1.282E 07	4.227E 05	1.324E 07	1.589E 04	1.228E 03	1.325E 06	1.228E 03	1.325E 06	1.458E 07
0.07	0.26	9.143E 06	3.425E 05	9.506E 06	1.344E 04	1.210E 03	1.167E 06	1.210E 03	1.167E 06	1.058E 07
0.08	0.30	6.672E 06	3.166E 05	6.989E 06	1.158E 04	1.228E 03	1.041E 06	1.228E 03	1.041E 06	8.042E 06
0.09	0.33	4.970E 06	2.806E 05	5.250E 06	1.014E 04	1.225E 03	9.347E 05	1.225E 03	9.347E 05	6.196E 06
0.10	0.37	3.777E 06	2.518E 05	4.029E 06	9.016E 03	1.220E 03	8.479E 05	1.220E 03	8.479E 05	4.889E 06
0.20	0.74	6.065E 06	1.205E 05	7.270E 05	4.464E 03	4.345E 05	4.345E 05	4.345E 05	4.345E 05	1.137E 06
0.30	1.41	2.316E 05	7.102E 04	3.029E 05	3.092E 03	2.507E 05	2.507E 05	2.507E 05	2.507E 05	5.573E 05
0.40	1.48	58	4.03E 04	1.655E 05	2.398E 03	1.764E 05	1.764E 05	1.764E 05	1.764E 05	3.447E 05
0.50	1.85	73	7.246E 04	1.007E 05	1.976E 03	1.511E 05	1.511E 05	1.511E 05	1.511E 05	2.388E 05
0.60	2.22	87	4.580E 04	1.874E 04	1.690E 03	1.690E 03	1.690E 03	1.690E 03	1.690E 03	1.728E 05
0.80	2.96	117	1.917E 04	9.140E 03	2.831E 04	1.324E 03	8.242E 04	4.094E 02	8.242E 04	1.125E 05
1.00	3.70	146	7.525E 03	4.916E 03	1.244E 04	1.397E 03	6.742E 04	4.334E 02	6.742E 04	1.129E 04
1.25	4.63	182	1.747E 03	2.306E 03	4.093E 03	9.096E 02	5.048E 02	2.479E 02	5.048E 02	6.166E 04
1.50	5.56	219	2.703E 02	9.927E 02	1.263E 03	4.823E 02	4.921E 04	2.149E 02	4.921E 04	6.147E 04
1.75	6.48	255	3.195E 01	3.787E 02	4.108E 02	4.897E 02	4.088E 04	1.836E 02	4.088E 04	4.537E 04
2.00	7.41	291	3.314E 01	1.253E 02	1.282E 02	6.179E 02	3.997E 04	1.549E 02	3.997E 04	4.083E 04
2.50	9.26	365	1.002E -02	7.802E 01	7.812E 01	5.177E 02	3.387E 04	1.142E 02	3.387E 04	3.451E 04
3.00	11.11	437	0.0	1.536E 01	1.536E 01	3.473E 02	2.467E 04	8.827E 01	2.467E 04	3.017E 04
3.50	12.96	510	0.0	1.026E -04	1.026E -04	3.944E 02	2.655E 04	6.441E 01	2.655E 04	2.791E 04
4.00	14.81	583	0.0	0.0	0.0	3.527E 02	2.415E 04	4.506E 01	2.415E 04	2.456E 04
4.50	16.67	656	0.0	0.0	0.0	3.186E 02	2.213E 04	4.407E 01	2.213E 04	2.249E 04
5.00	18.52	729	0.0	0.0	0.0	2.901E 02	2.035E 04	3.654E 01	2.035E 04	2.072E 04
6.00	22.22	875	0.0	0.0	0.0	2.449E 02	1.757E 04	2.541E 01	1.757E 04	1.780E 04
8.00	29.23	1167	0.0	0.0	0.0	1.828E 02	1.346E 04	1.531E 01	1.346E 04	1.365E 04
10.00	37.04	1458	0.0	0.0	0.0	1.413E 02	1.073E 04	7.395E 00	1.073E 04	1.088E 04

\* ELECTRON MODELS:  
 A66: INNER ZONE-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 SOLPRO: SOLAR FLARE PROTONS AT 1 AU  
 UNLIMITED TIME IN LAMINAR FLOW  
 FOR CUTOFF DIPOLE SHELL OF 5 F.P.  
 FOR TRAP MODE 0=007: 3 OF AL EVENTS=6  
 66.8% OF MAGNETIC SHIELDING APPLIED

NOTE: 0 DENOTES THE DEGREE OF CONFIDENCE (NF WISHES TO ASSIGN TO RESULTS, NAMELY THAT FOR THE SPECIFIC MISSION DURATION THE CALCULATED FLUENCES ARE THE SMALLEST VALUES WHICH WILL NOT BE EXCEEDED BY ACTUALLY ENCOUNTERED INTENSITIES)

\*\* PROTON MODEL:  
 A6-MAC: TRAPPED PROTONS-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR PROTON SPECTRA NEITHER TOWARDS LOWER NOR TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS USED IN THE CONSTRUCTION OF THE MODEL (SATEL-LITE MEASUREMENTS MADE DURING THE 20TH SOLAR CYCLE: 1964-1975) DO NOT CONTAIN INFORMATION FOR F<10 AND F>200 KEV.







\*\*\*\*\*  
 \*\* JORBITA BULK STUDY WITH COMPOSITE PARTICLE INFLUENCE: VEIES AFB; AEL7 FOR 30.0N MAXIMUM  
 \*\* UNCERTAINTY FACTORS (U1) APPLIED FOR THIS RUN ARE: FCF PRCTNS (AFH) - JFE 1.0; FCF INFLU ZONE  
 \*\* ANGELIC COORDINATES 3 AND 4 COMPUTED BY INAPA CF 1972 WITH ALL MAG. MODEL 4; FABRICATION BY ATAL 16-76-1576; TIME 1980.5  
 \*\* ANGELIC COORDINATES 5 AND 6 COMPUTED BY INAPA CF 1972 WITH ALL MAG. MODEL 4; FABRICATION BY ATAL 16-76-1576; TIME 1980.5  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT S.G. STASINCOULOS AT NASA-CSE/C/OLE 601; GAITHERLY, MARYLAND 20771; TEL: 301-344-8067  
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 \*\* MODEL: DESE IN SEMI-INFINITE ALUMINUM MEDIUM \*\*\*\*\*  
 \*\* MISSION DURATION: 5.0000 YEARS \*\*\*\*\*  
 \*\*\*\*\*

SHIELD THICKNESS (ALUMINUM)	T		INNER ZONE		ELECTRICITY		CURRENTS - AMPLUNG		TRAPEZOIDAL CURRENTS	RECTANGULAR		TOTAL CURRENTS
	IN	OUT	IN	OUT	IN	OUT	IN	OUT		IN	OUT	
0.01	6.04	1.0	1.782E	CF	7.653E	06	1.359E	08	1.158E	CF	3.018E	08
0.02	6.07	3.0	5.258E	CF	8.716E	06	7.703E	07	4.667E	CF	1.884E	08
0.03	0.11	4.0	5.551E	CF	3.403E	06	5.703E	07	2.512E	CF	1.407E	07
0.04	0.15	6.0	3.556E	CF	2.687E	06	2.225E	07	1.584E	CF	5.411E	07
0.05	0.19	7.0	2.387E	CF	2.157E	06	2.225E	07	1.072E	CF	2.678E	07
0.06	0.22	9.0	1.652E	CF	1.814E	06	1.832E	07	7.077E	CF	2.612E	07
0.07	0.26	10.0	1.130E	CF	1.605E	06	1.232E	07	6.444E	CF	1.810E	07
0.08	0.20	12.0	8.438E	CF	1.367E	06	9.803E	06	5.272E	CF	4.525E	06
0.09	0.23	13.0	6.145E	CF	1.214E	06	7.375E	06	3.527E	CF	1.435E	07
0.10	0.27	15.0	4.538E	CF	1.130E	06	5.845E	06	2.920E	CF	1.566E	06
0.20	0.74	29.0	3.132E	CF	5.154E	05	4.322E	05	2.029E	CF	2.827E	06
0.30	1.11	44.0	4.047E	CF	3.022E	05	3.311E	07	1.667E	CF	1.523E	06
0.40	1.48	58.0	1.648E	CF	1.848E	05	1.957E	05	1.067E	CF	2.682E	05
0.50	1.85	73.0	4.165E	CF	1.118E	05	2.755E	03	7.240E	CF	1.112E	05
0.60	2.22	87.0	1.650E	CF	2.667E	04	2.869E	04	4.444E	CF	1.556E	05
0.80	2.96	117.0	4.785E	CF	3.647E	04	3.645E	04	1.068E	CF	1.304E	05
1.00	3.70	146.0	1.253E	CF	1.532E	04	1.945E	04	7.523E	CF	6.411E	04
1.25	4.62	182.0	2.720E	CF	8.578E	03	2.977E	03	4.867E	CF	1.861E	04
1.50	5.56	219.0	3.747E	CF	3.859E	03	3.863E	03	3.245E	CF	1.602E	04
1.75	6.48	255.0	5.457E	CF	1.477E	03	4.601E	03	2.200E	CF	1.202E	04
2.00	7.41	292.0	5.522E	CF	3.184E	02	4.885E	02	1.561E	CF	7.765E	03
2.50	9.26	365.0	3.0	3.0	2.668E	01	2.668E	01	7.824E	CF	5.471E	03
3.00	11.11	437.0	3.0	3.0	5.556E	01	5.048E	01	4.186E	CF	2.461E	03
3.50	12.96	510.0	3.0	3.0	5.275E	04	3.0	3.0	1.813E	CF	2.464E	03
4.00	14.81	583.0	3.0	3.0	3.0	3.0	3.0	3.0	1.153E	CF	2.108E	03
4.50	16.67	656.0	3.0	3.0	3.0	3.0	3.0	3.0	6.655E	CF	1.255E	03
5.00	18.52	729.0	3.0	3.0	3.0	3.0	3.0	3.0	4.463E	CF	1.774E	03
6.00	22.82	873.0	3.0	3.0	3.0	3.0	3.0	3.0	3.344E	CF	1.517E	03
8.00	29.63	1167.0	3.0	3.0	3.0	3.0	3.0	3.0	2.847E	CF	1.318E	03
10.00	37.04	1459.0	3.0	3.0	3.0	3.0	3.0	3.0	2.555E	CF	4.614E	02
									1.922E	CF	4.664E	02

\* ELECTRON MODEL:  
 AEG: INNER ZONE SOLAR MAX  
 NO UNCERTAINTY FACTOR HAS BEEN ADDED TO THE MODEL DATA.

AEL7: OUTER ZONE-INTERIOR MODEL WITHOUT SOLAR CYCLE DEPENDENCE.  
 FOR ENERGIES ABOVE 1.5 KEV, THE MODEL CONTAINS UPPER AND  
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN  
 EXISTING DATA SETS. THE BEST FIT FAVORS VARIATIONS FIT  
 TO OUTER ZONE DATA. ALL DATA SETS ARE AVAILABLE FOR REUSE.  
 ALL THE DATA SETS ORIGINALLY AVAILABLE REPRESENTATIVE U  
 \*\* THE BETTER VERSION WAS USED FOR THESE CALCULATIONS \*\*

\*\* PRTION MODEL:  
 AEG: INNER ZONE SOLAR MAX  
 NO UNCERTAINTY FACTOR HAS BEEN ADDED TO THE MODEL DATA.

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR  
 SPECTRA EITHER TOWARDS LOWER OR  
 TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS  
 USED IN THE CONSTRUCTION OF THE MODEL LABEL  
 CYCLES (1944-1975) DO NOT CONTAIN INFORMATION  
 FOR ENERGIES ABOVE 1520 CFV.





ORBITAL FLUX STUDY WITH COMPOSITE PARTICLE ENVIRONMENTS: RESULTS ARE AVAILABLE FOR SOLAR MAXIMUM AND MINIMUM FLUXES OF 1979  
 UNCERTAINTY FACTORS (UF) APPLIED FOR THIS MODEL CONTAIN CERTAIN DEPENDENCE ON ENERGY AND TIME. THE MODEL IS BASED ON DATA FROM  
 MAGNETIC COORDINATES 0 AND 1 COMPUTED BY INVAR OF 1972. THE MODEL IS BASED ON DATA FROM 1979. THE MODEL IS BASED ON DATA FROM  
 VEHICLE. MAXIMUM MAXIMUM MAGNETIC COORDINATE 0 AND 1 COMPUTED BY INVAR OF 1972. THE MODEL IS BASED ON DATA FROM 1979. THE MODEL IS BASED ON DATA FROM  
 FOR INFORMATION OR EXPLANATION CONTACT E. J. STASINGROULIS AT NASA-355 CODE 601. GLENBELT 1, MARLANG 20771. TEL. (301)-344-8067

SHIELD THICKNESS (ALUMINUM)	ELECTRONS*				PROTONS				TOTAL DOSE	
	INNER ZN (RADS-ALL)	CUTER ZN (RADS-ALL)	TOTAL (RADS-ALL)	TRAPPED** (RADS-ALL)	SOLAR+ (RADS-ALL)	TOTAL (RADS-ALL)	ALL SOURCES (RADS-ALL)			
0.01	0.04	1.0	4.095E 08	5.961E 06	4.095E 08	1.399E 05	1.448E 07	2.311E 02	1.848E 07	4.281E 08
0.02	0.07	3.0	2.754E 06	3.980E 06	2.754E 06	1.259E 05	9.375E 06	2.335E 02	9.581E 06	2.289E 08
0.03	0.11	4.0	1.543E 09	2.904E 06	1.543E 09	1.061E 05	7.153E 06	2.355E 02	7.153E 06	2.046E 08
0.04	0.15	6.0	1.423E 08	2.300E 06	1.423E 08	8.015E 04	5.078E 06	2.379E 02	5.511E 06	1.905E 08
0.05	0.19	7.0	1.055E 08	1.978E 06	1.055E 08	7.401E 04	5.151E 06	2.404E 02	5.151E 06	1.127E 08
0.06	0.22	9.0	7.924E 07	1.680E 06	8.092E 07	6.553E 04	4.704E 06	2.421E 02	4.704E 06	8.569E 07
0.07	0.26	1.0	6.006E 07	1.455E 06	6.161E 07	5.462E 04	4.283E 06	2.456E 02	4.283E 06	6.586E 07
0.08	0.30	12.0	4.588E 07	1.279E 06	4.716E 07	4.920E 04	3.924E 06	2.493E 02	3.924E 06	5.114E 07
0.09	0.33	13.0	3.540E 07	1.138E 06	3.654E 07	4.310E 04	3.914E 06	2.522E 02	3.616E 06	4.320E 07
0.10	0.37	15.0	2.747E 07	1.021E 06	2.889E 07	3.799E 04	3.714E 06	2.550E 02	3.714E 06	3.189E 07
0.20	0.74	29.0	1.520E 06	5.213E 05	1.726E 06	1.726E 04	1.736E 06	3.010E 02	1.736E 06	6.527E 06
0.30	1.11	44.0	1.518E 06	3.536E 05	1.871E 06	1.688E 04	1.110E 06	2.650E 02	1.110E 06	2.996E 06
0.40	1.48	58.0	7.878E 05	2.504E 05	1.038E 06	8.995E 03	7.811E 05	2.288E 02	7.834E 05	1.431E 06
0.50	1.85	73.0	4.919E 05	1.772E 05	6.691E 05	7.353E 03	5.772E 05	2.046E 02	5.772E 05	1.256E 06
0.60	2.22	87.0	3.351E 05	1.257E 05	4.608E 05	6.372E 03	4.608E 05	1.813E 02	4.623E 05	9.293E 05
0.80	2.96	117.0	1.729E 05	6.633E 04	2.392E 05	4.887E 03	3.248E 05	1.527E 02	3.263E 05	5.704E 05
1.00	3.70	146.0	9.161E 04	3.920E 04	1.308E 05	4.038E 03	2.479E 05	1.302E 02	2.452E 05	3.840E 05
1.25	4.93	182.0	3.203E 04	2.279E 04	5.582E 04	3.334E 03	1.998E 05	1.100E 02	2.006E 05	2.596E 05
1.50	5.56	219.0	7.180E 03	1.275E 04	1.993E 04	2.852E 03	1.596E 05	9.264E 02	1.705E 05	1.933E 05
1.75	6.48	255.0	1.031E 03	6.166E 03	7.197E 03	2.501E 03	1.521E 05	8.213E 02	1.520E 05	1.626E 05
2.00	7.41	292.0	1.375E 02	2.694E 03	2.831E 03	2.235E 03	1.354E 05	7.234E 02	1.421E 05	1.452E 05
2.50	9.26	365.0	5.766E -01	2.470E 02	2.475E 02	1.863E 03	1.176E 05	5.345E 02	1.176E 05	1.202E 05
3.00	11.11	417.0	0.0	1.053E -01	1.053E 01	1.441E 03	1.321E 05	4.523E 02	1.026E 05	1.042E 05
3.50	12.96	510.0	0.0	1.543E -02	1.543E -02	1.441E 03	9.950E 04	3.021E 02	5.066E 04	9.230E 04
4.00	14.81	583.0	0.0	0.0	0.0	1.305E 03	8.326E 04	2.948E 02	8.326E 04	8.486E 04
4.50	16.67	656.0	0.0	0.0	0.0	1.195E 03	7.795E 04	2.555E 02	7.795E 04	7.940E 04
5.00	18.52	729.0	0.0	0.0	0.0	1.114E 03	7.316E 04	2.146E 02	7.316E 04	7.447E 04
5.50	20.38	802.0	0.0	0.0	0.0	9.584E 02	6.533E 04	1.568E 02	6.545E 04	6.845E 04
6.00	22.22	875.0	0.0	0.0	0.0	7.635E 02	5.932E 04	1.132E 02	5.932E 04	5.399E 04
10.00	37.04	1458.0	0.0	0.0	0.0	4.467E 02	3.117E 04	3.322E 01	3.120E 04	3.165E 04

\* ELECTRON MODELS:  
 AF6: INNER ZONE-SOLAR MAX  
 NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 SOLPRO: SOLAR FLARE PROTONS AT 1 AU  
 LONATIENALIA, INDEPENDENTARY  
 FOR CURIFF DIPOLE SHELL OF 5 E.R.  
 \*\* PROTON MODEL:  
 AP6: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE.  
 FOR ENERGIES ABOVE 1.5 MEV THIS MODEL CONTAINS CERTAIN DEPENDENCE.  
 LOWER LIMIT VALUES TO ACCOUNT FOR DISCREPANCY BETWEEN  
 EXISTING DATA SETS. THE AE17-H1 FAVORS AMPLETT'S FIT  
 TO 0.01-1.0 DATA WHILE AE17-L0 IS MORE REPRESENTATIVE.  
 ALL THE DATA SETS PRESENTLY AVAILABLE TO ASSDC.  
 \*\* THE AE17-0 WRSTON WAS USED FOR THESE CALCULATIONS \*\*

IT IS NOT ADVISABLE TO EXTRAPOLATE THE SOLAR  
 PROTON SPECTRA TO ENERGIES LOWER THAN  
 TOWARDS HIGHER ENERGIES BECAUSE THE DATA SETS  
 USED IN THE CALCULATION OF THE MODEL LABEL  
 LITE WFA SUBELEMENTS MADE DURING THE SOLAR  
 CYCLE 1964-1974 DO NOT CONTAIN INFORMATION  
 FOR 4-10 AND 4-20 MEV

TABLE 56



ORBITAL FLUX SLOPE WITH COMPOSITE PARTICLE FRAGMENTATION VILTES AP01 AL17 FOR SOLAR MAXIMUM...  
 \*\* UNCERTAINTY FACTORS (UP) APPLIED FOR THIS AREA: 1.00 FOR INNER ZONE, 1.00 FOR UPPER ZONE, 1.00 FOR LOWER ZONE...  
 \*\* MAGNETIC COORDINATES B AND L COMPLETED BY INVATA CF 1572 WITH ALLMAG MODEL 4: EARTH MAGNETIC FIELD...  
 \*\* INFORMATION OR EXPLANATION CONTACT 488 INCLINATION CODES...  
 \*\* FOR INFORMATION OR EXPLANATION CONTACT 488 INCLINATION CODES...  
 \*\* MISSION DURATION: 5.0000 YEAR(S) \*\*\*

SHIELD THICKNESS (G-CM)	ELECTRONS* (G-CM-AL)	TOTAL ALUMINUM (G-CM-AL)	ELECTRONS*			TOTAL ALUMINUM (G-CM-AL)	TOTAL ELECTRONS (G-CM-AL)
			INNER ZONE (G-CM-AL)	CUTOFF ZONE (G-CM-AL)	TOTAL (G-CM-AL)		
0.01	0.04	1.778E 07	7.467E 07	2.977E 05	2.763E 05	1.00E 07	
0.02	0.07	1.578E 07	5.101E 06	1.908E 05	1.908E 05	6.0E 06	
0.03	0.11	5.936E 06	2.531E 06	1.532E 05	1.532E 05	2.0E 06	
0.04	0.14	4.156E 06	1.823E 06	1.262E 05	1.262E 05	1.5E 06	
0.05	0.19	3.029E 06	1.421E 06	9.513E 04	9.513E 04	1.1E 06	
0.06	0.22	2.192E 06	1.019E 06	7.004E 04	7.004E 04	8.0E 05	
0.07	0.26	1.583E 06	7.481E 05	5.255E 04	5.255E 04	6.0E 05	
0.08	0.30	1.145E 06	5.415E 05	3.919E 04	3.919E 04	4.5E 05	
0.09	0.33	8.192E 05	3.915E 05	2.899E 04	2.899E 04	3.3E 05	
0.10	0.37	6.024E 05	2.818E 05	2.115E 04	2.115E 04	2.4E 05	
0.20	0.74	3.874E 05	1.808E 05	1.370E 04	1.370E 04	1.5E 05	
0.30	1.11	2.574E 05	1.191E 05	9.071E 03	9.071E 03	1.0E 05	
0.40	1.48	1.743E 05	8.372E 04	6.442E 03	6.442E 03	7.5E 04	
0.50	1.85	1.190E 05	5.832E 04	4.352E 03	4.352E 03	5.0E 04	
0.60	2.22	8.244E 04	4.244E 04	3.155E 03	3.155E 03	3.6E 04	
0.70	2.59	5.874E 04	3.071E 04	2.285E 03	2.285E 03	2.7E 04	
0.80	2.96	4.241E 04	2.220E 04	1.680E 03	1.680E 03	1.9E 04	
1.00	3.76	2.752E 04	1.460E 04	1.098E 03	1.098E 03	1.2E 04	
1.20	4.56	1.905E 04	1.049E 04	7.929E 02	7.929E 02	9.0E 03	
1.50	5.76	1.149E 04	6.844E 03	5.282E 02	5.282E 02	6.0E 03	
2.00	7.56	6.402E 03	3.922E 03	2.931E 02	2.931E 02	3.3E 03	
3.00	1.12	3.411E 03	2.107E 03	1.599E 02	1.599E 02	1.8E 03	
4.00	1.48	2.301E 03	1.441E 03	1.093E 02	1.093E 02	1.2E 03	
5.00	1.84	1.621E 03	1.021E 03	7.871E 01	7.871E 01	9.0E 02	
6.00	2.20	1.151E 03	7.351E 02	5.691E 01	5.691E 01	6.5E 02	
8.00	2.96	6.741E 02	4.441E 02	3.351E 01	3.351E 01	3.8E 02	
10.00	3.76	4.331E 02	2.871E 02	2.171E 01	2.171E 01	2.5E 02	

\* ELECTRON MODELS:  
 A66: INNER ZONE-SOLAR MAX. NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 A67: OUTER ZONE-INTERIM MODEL WITHOUT SOLAR CYCLE DEPENDENCE. FOR ENERGIES ABOVE 15 KEV, THIS MODEL CONTAINS DEPENDENCE ON MAGNETIC COORDINATES TO ACCOUNT FOR DISCREPANCY BETWEEN EXISTING DATA SETS. THE A67-HI-ENERGY FACTORS ARE APPLIED TO THE DATA WHILE THE A67-LO-ENERGY IS IN GENERAL REPRESENTATIVE OF ALL THE DATA SETS PRESENTLY AVAILABLE TO NSSDC.  
 \*\* THE A67-LO-ENERGY WAS USED FOR THESE CALCULATIONS \*\*  
 \*\* PROTON MODEL:  
 A68-MAC: ISAREC PROTONS-SOLAR MAX. NO UNCERTAINTY FACTOR WAS APPLIED TO THE MODEL DATA.  
 \*\* SOLAR PERTINENCE:  
 SOURCE: SOLAR FLUX MEASUREMENTS (NATIONAL BUREAU OF STANDARDS)  
 THE DATA SETS WERE OBTAINED FROM THE SOLAR WIND OBSERVATORY AT LAS CRUCES, NEW MEXICO, U.S.A.  
 FOR THE CYCLES 1957-58 AND 1958-59.  
 \*\* CYCLE DEPENDENCE \*\*  
 \*\* PROTONS \*\*  
 NOTE: THIS MODEL IS BASED ON THE DATA SETS FROM THE LAS CRUCES OBSERVATORY. THE CALCULATED VALUES WILL NOT BE ACCURATE FOR OTHER OBSERVATIONS OR FOR OTHER CYCLES.  
 \*\* DATA SETS \*\*  
 THE DATA SETS CONTAIN INFORMATION ON THE CYCLE DEPENDENCE OF THE SOLAR FLUX MEASUREMENTS. THE DATA SETS USED IN THIS ANALYSIS ARE THE LAS CRUCES DATA SETS (SOLAR WIND OBSERVATORY) AND THE LAS CRUCES DATA SETS (SOLAR WIND OBSERVATORY). THE DATA SETS WILL CONTAIN INFORMATION ON THE CYCLE DEPENDENCE OF THE SOLAR FLUX MEASUREMENTS.  
 \*\* DATA SETS \*\*  
 THE DATA SETS CONTAIN INFORMATION ON THE CYCLE DEPENDENCE OF THE SOLAR FLUX MEASUREMENTS. THE DATA SETS USED IN THIS ANALYSIS ARE THE LAS CRUCES DATA SETS (SOLAR WIND OBSERVATORY) AND THE LAS CRUCES DATA SETS (SOLAR WIND OBSERVATORY). THE DATA SETS WILL CONTAIN INFORMATION ON THE CYCLE DEPENDENCE OF THE SOLAR FLUX MEASUREMENTS.

TABLE 5B



\*\*\*\*\* UNIFLX CF 1979 \*\*\*\*\*  
 \*\* ORBITAL FLUX ENERGY WILL BE USED IN THE MODEL. ELECTRONIC (AE) - CF 1.0 \*\*  
 \*\* UNCERTAINTY FACTORS OF 10% WERE USED FOR THE MODEL. UNIFORMITY OF THE DATA SETS \*\*  
 \*\* MAGNETIC COEFFICIENTS 3 AND 4 WERE USED IN THE MODEL. UNIFORMITY OF THE DATA SETS \*\*  
 \*\* FOR INFORMATION ON THE MODEL CONTACT THE AUTHOR AT THE ADDRESS ABOVE. \*\*  
 \*\* \*\* \*\* \*\*

TABLE NUMBER	TABLE NAME	TABLE TYPE	TABLE SIZE	TABLE DATE	TABLE TIME	TABLE UNIT	TABLE VALUE	TABLE ERROR	TABLE TOTAL	TABLE SOURCE
0.01	0.04	1.0	0.0000	07	1.0000	00	0.0000	0.0000	0.0000	0.0000
0.02	0.07	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.03	0.11	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.04	0.15	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.05	0.20	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.06	0.25	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.07	0.30	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.08	0.35	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.09	0.40	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.10	0.45	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.11	0.50	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.12	0.55	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.13	0.60	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.14	0.65	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.15	0.70	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.16	0.75	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.17	0.80	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.18	0.85	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.19	0.90	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.20	0.95	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000
0.21	1.00	1.0	0.0000	07	0.0000	07	0.0000	0.0000	0.0000	0.0000

\*\*\*\*\* UNIFLX CF 1979 \*\*\*\*\*  
 \*\* ORBITAL FLUX ENERGY WILL BE USED IN THE MODEL. ELECTRONIC (AE) - CF 1.0 \*\*  
 \*\* UNCERTAINTY FACTORS OF 10% WERE USED FOR THE MODEL. UNIFORMITY OF THE DATA SETS \*\*  
 \*\* MAGNETIC COEFFICIENTS 3 AND 4 WERE USED IN THE MODEL. UNIFORMITY OF THE DATA SETS \*\*  
 \*\* FOR INFORMATION ON THE MODEL CONTACT THE AUTHOR AT THE ADDRESS ABOVE. \*\*  
 \*\* \*\* \*\*

\*\*\*\*\* UNIFLX CF 1979 \*\*\*\*\*  
 \*\* ORBITAL FLUX ENERGY WILL BE USED IN THE MODEL. ELECTRONIC (AE) - CF 1.0 \*\*  
 \*\* UNCERTAINTY FACTORS OF 10% WERE USED FOR THE MODEL. UNIFORMITY OF THE DATA SETS \*\*  
 \*\* MAGNETIC COEFFICIENTS 3 AND 4 WERE USED IN THE MODEL. UNIFORMITY OF THE DATA SETS \*\*  
 \*\* FOR INFORMATION ON THE MODEL CONTACT THE AUTHOR AT THE ADDRESS ABOVE. \*\*  
 \*\* \*\* \*\*



AMBIENT THERMOSPHERIC ENVIRONMENT: NVLX1: S-MAX 600E:UR 1667KM 1667KM FOR SOL MAX

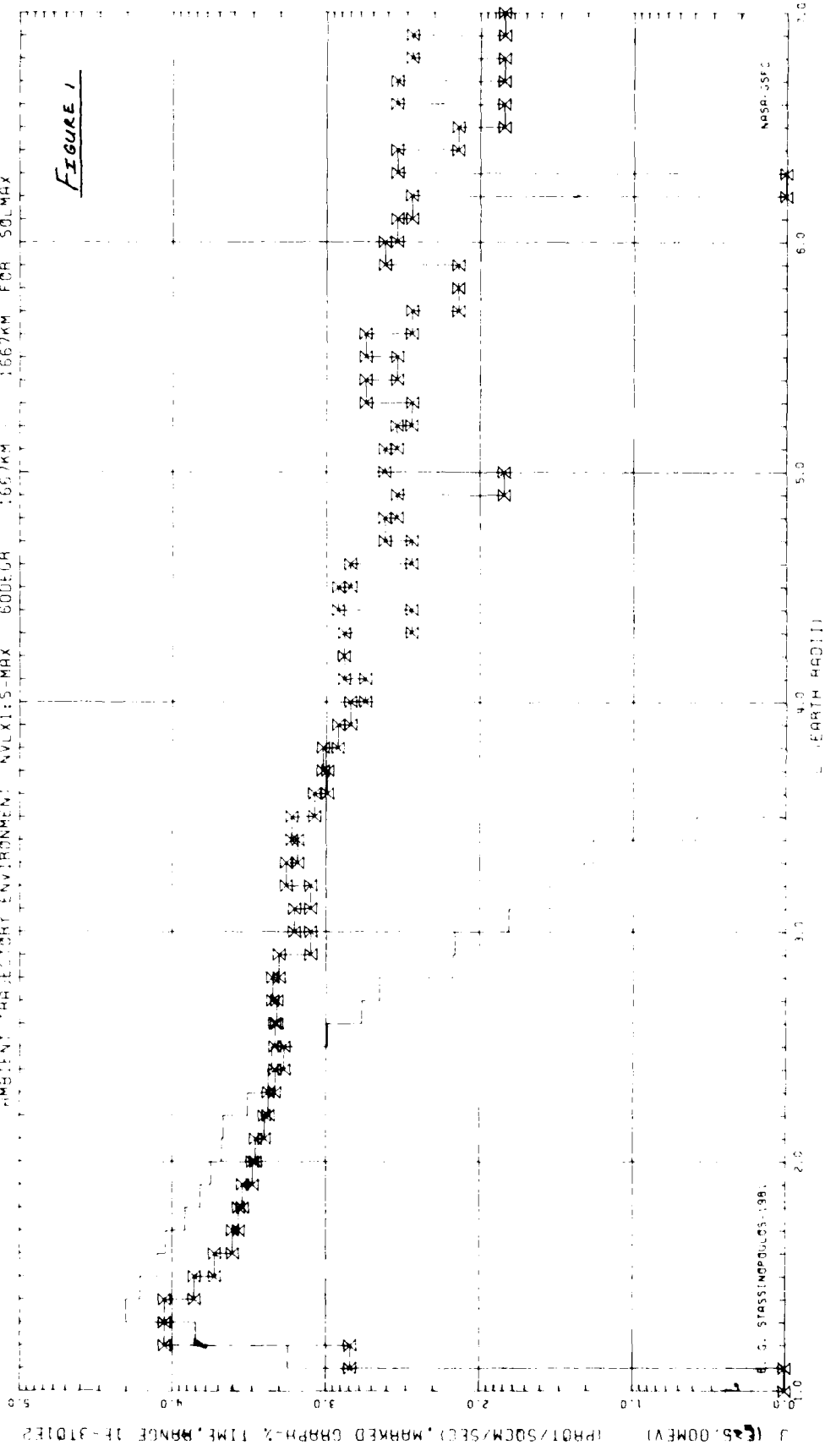


FIGURE 1

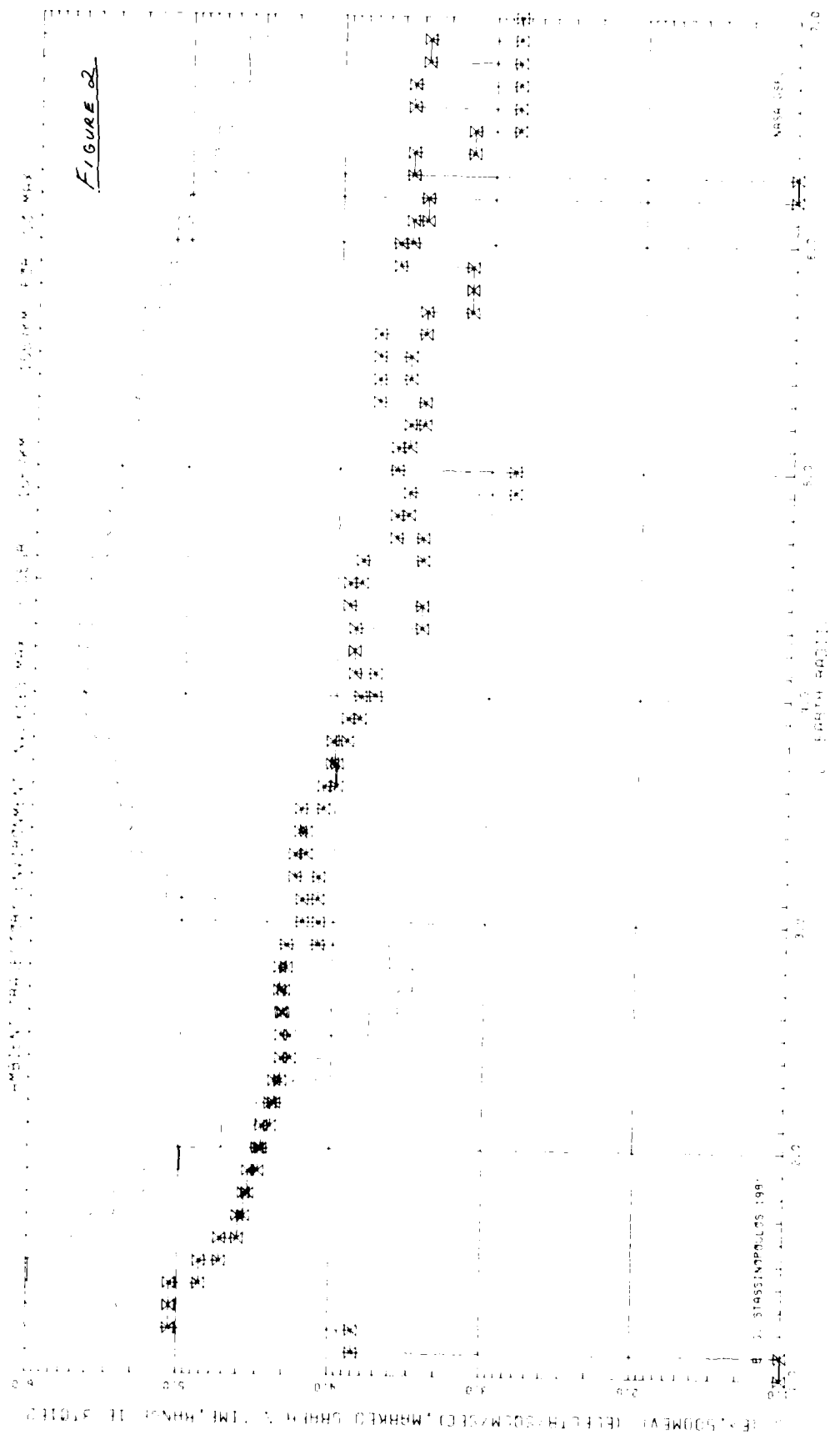


FIGURE 2

1000 HAZEL

STASSIN/PLDS 199

MARSH USA

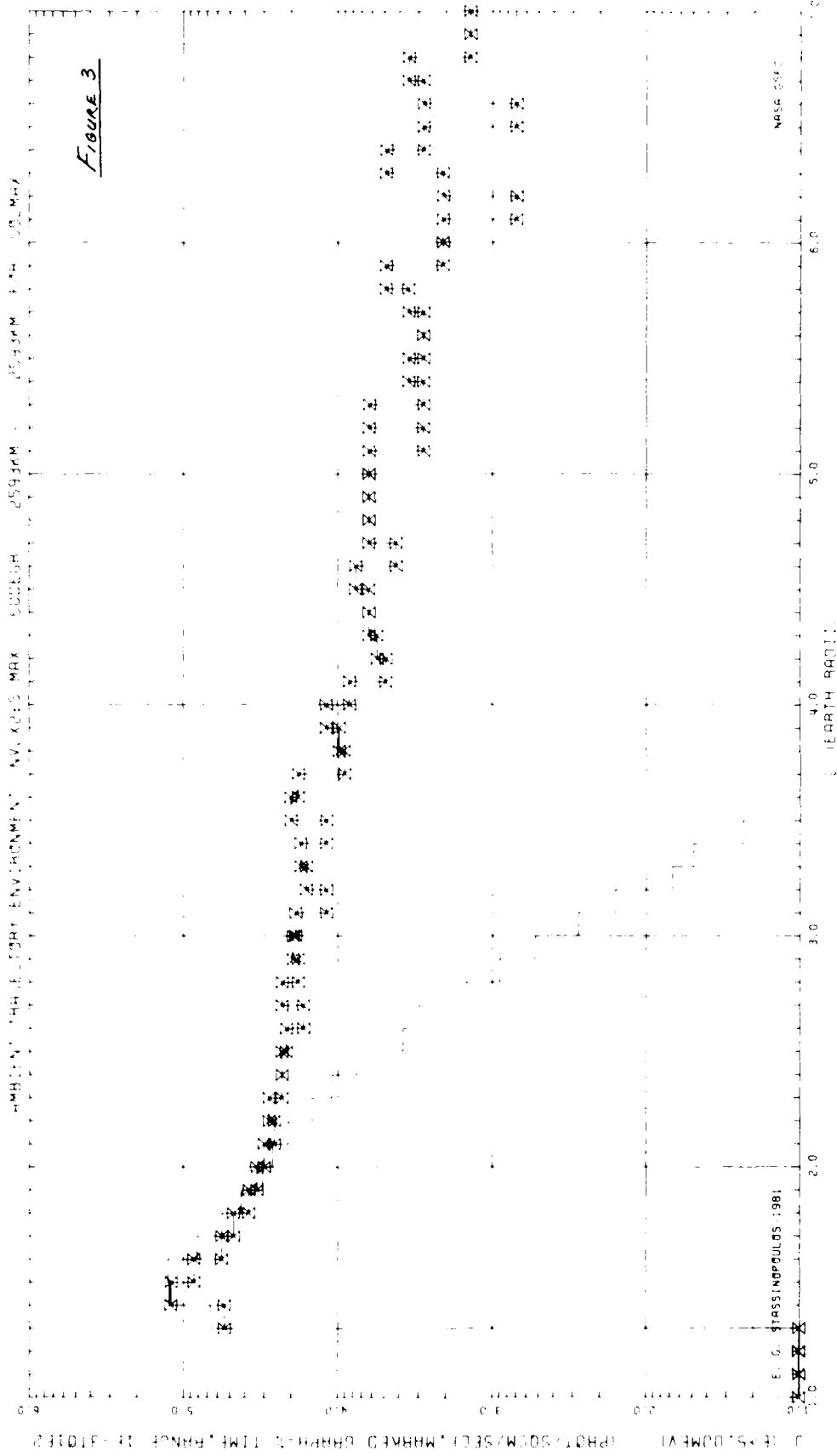
1000 HAZEL

ELEVATION (FEET) (SOM/SECT), MARKED ORIGIN IN TIME, RANGE IN FEET

1000 HAZEL

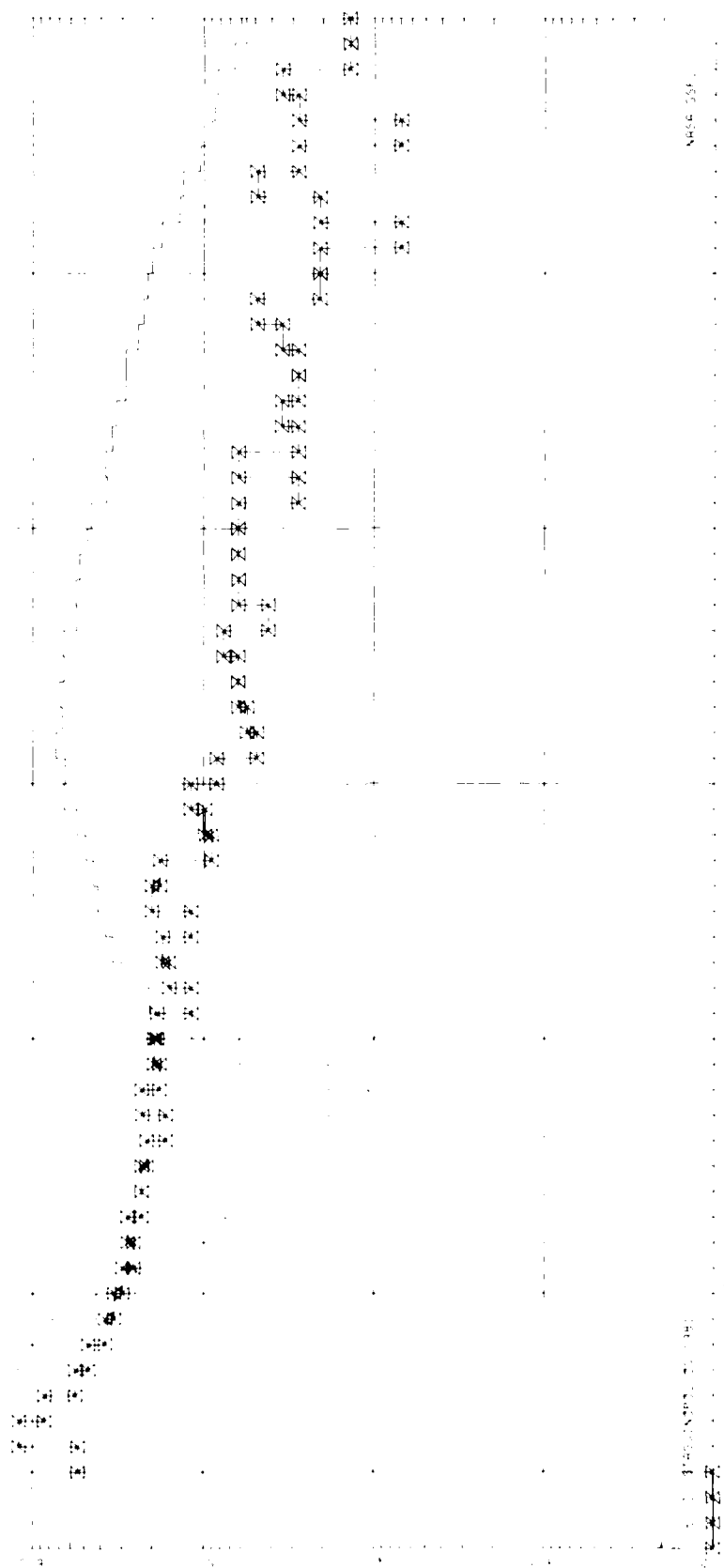
AMBIENT THERMAL ENVIRONMENT NY, K215 MAX 50000R 2590PM 14H 30 MAY

FIGURE 3



AMOUNT TO BE PROVIDED BY THE GOVERNMENT IN THE YEAR 2000 2001 2002 2003 2004 2005 2006 2007 2008 2009 2010 2011 2012 2013 2014 2015 2016 2017 2018 2019 2020 2021 2022 2023 2024 2025 2026 2027 2028 2029 2030 2031 2032 2033 2034 2035 2036 2037 2038 2039 2040 2041 2042 2043 2044 2045 2046 2047 2048 2049 2050 2051 2052 2053 2054 2055 2056 2057 2058 2059 2060 2061 2062 2063 2064 2065 2066 2067 2068 2069 2070 2071 2072 2073 2074 2075 2076 2077 2078 2079 2080 2081 2082 2083 2084 2085 2086 2087 2088 2089 2090 2091 2092 2093 2094 2095 2096 2097 2098 2099 2100

FIGURE 4

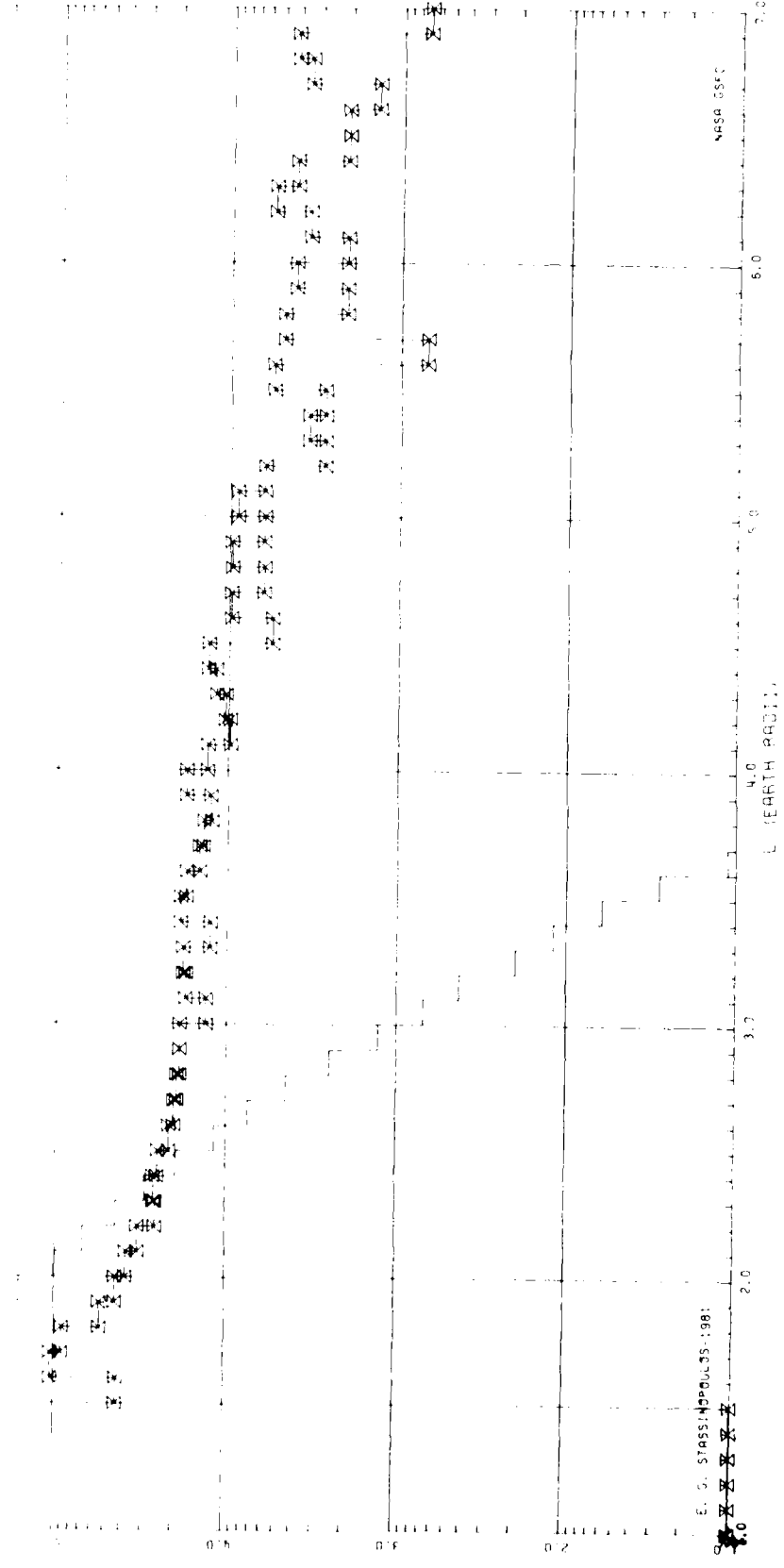


NRSE 554

2000 2001 2002 2003 2004 2005 2006 2007 2008 2009 2010 2011 2012 2013 2014 2015 2016 2017 2018 2019 2020 2021 2022 2023 2024 2025 2026 2027 2028 2029 2030 2031 2032 2033 2034 2035 2036 2037 2038 2039 2040 2041 2042 2043 2044 2045 2046 2047 2048 2049 2050 2051 2052 2053 2054 2055 2056 2057 2058 2059 2060 2061 2062 2063 2064 2065 2066 2067 2068 2069 2070 2071 2072 2073 2074 2075 2076 2077 2078 2079 2080 2081 2082 2083 2084 2085 2086 2087 2088 2089 2090 2091 2092 2093 2094 2095 2096 2097 2098 2099 2100

MEASUREMENTS OF THE ELECTRIC FIELD IN THE MANTLE

FIGURE 5



1 (E > 5.0 MEV) (PROT/SEC/SEC) (MHR) 2 (TIME RANGE) 3 (DATE)

AMBIENT AIR QUALITY ENVIRONMENT, NEWARK, N.J., 1981

3885PM 3885PM 3885PM

0305HR

0305HR

0305HR

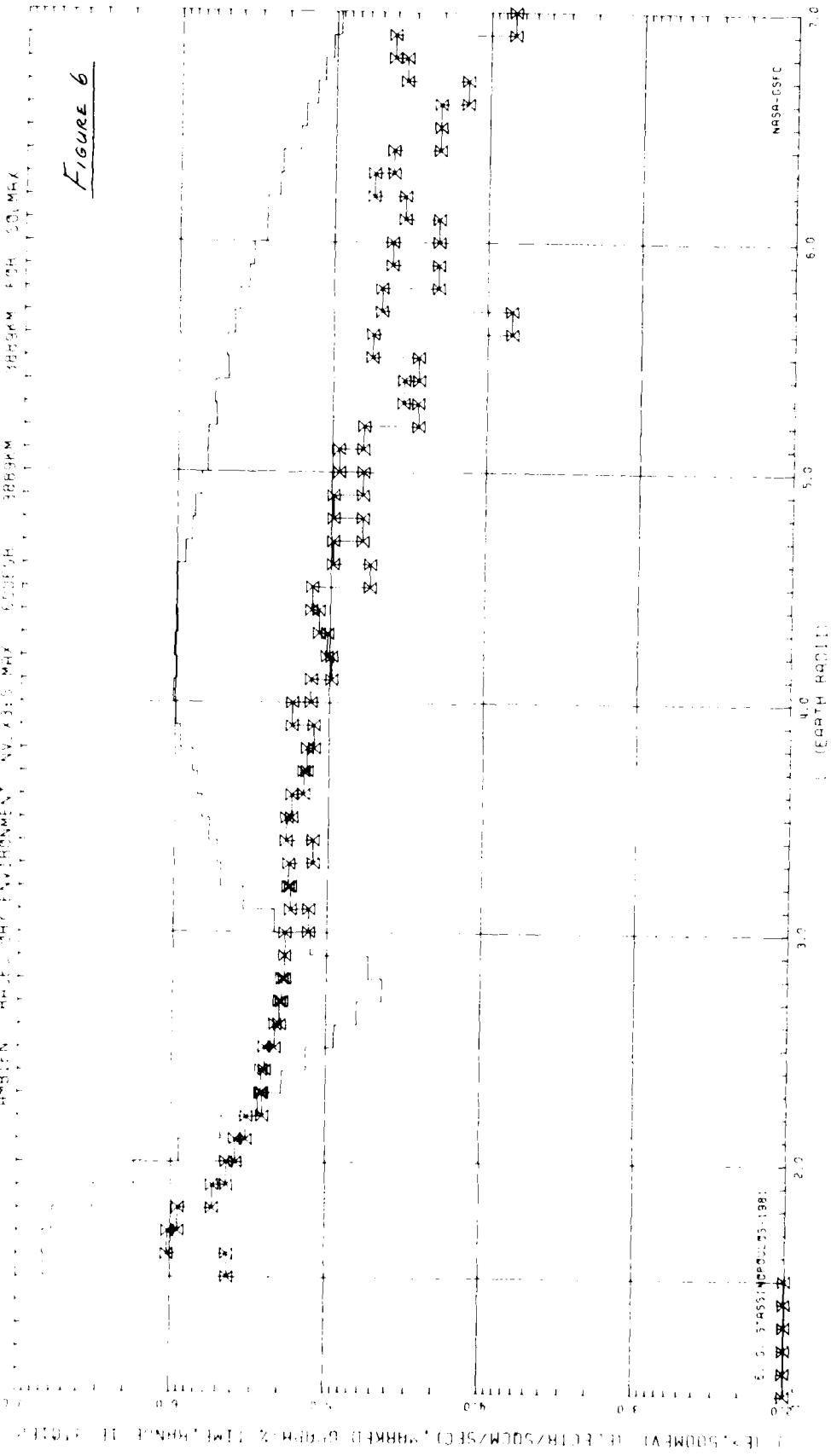
0305HR

0305HR

0305HR

0305HR

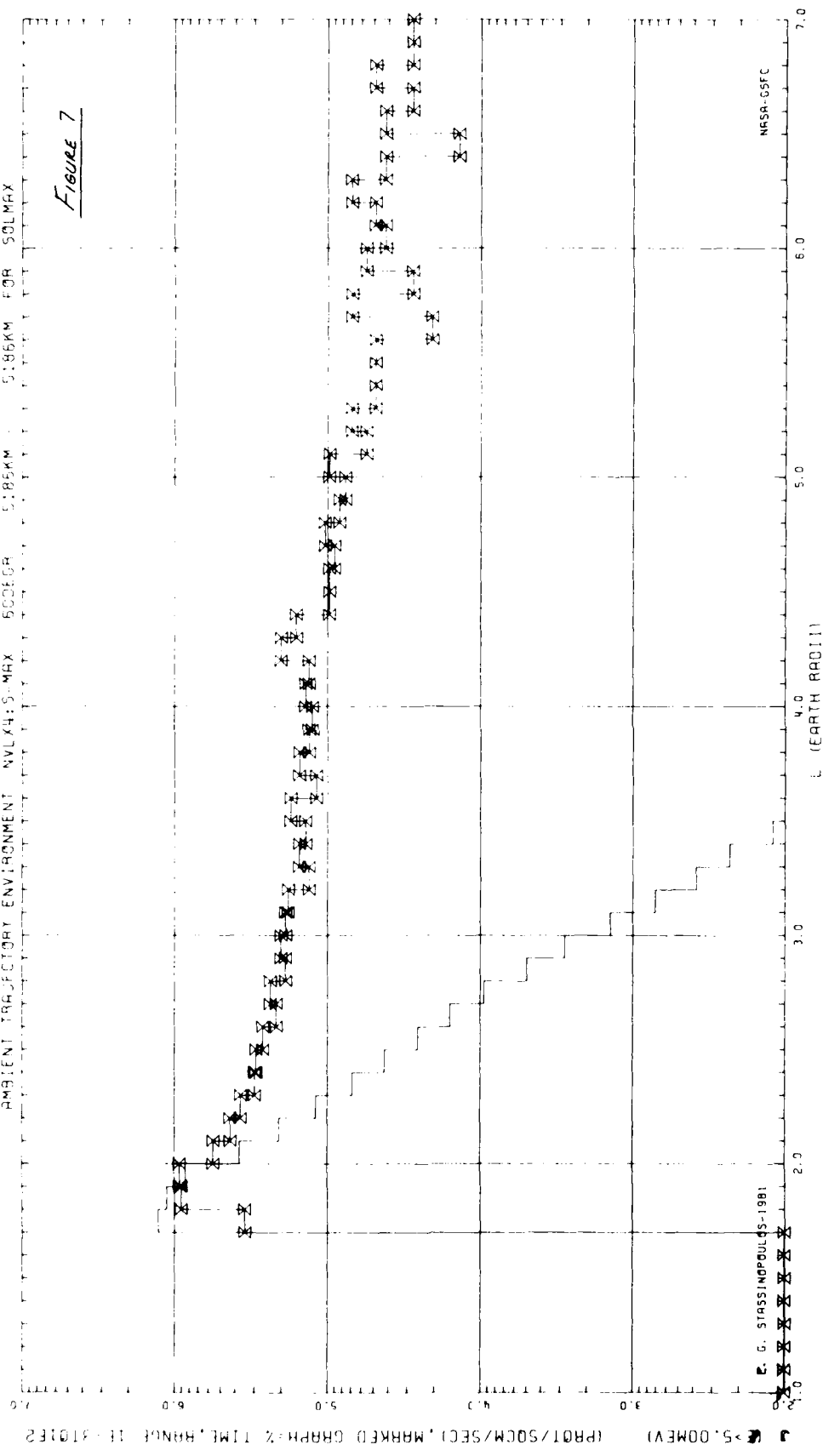
Figure 6



U.S. STRESS: MPO-02-73-1981

AMBIENT TRAJECTORY ENVIRONMENT NUL4: S-MAX 5006CR 5185KM 5185KM FOR SOLMAX

Figure 7



E. G. STASSINPOULOS-1981

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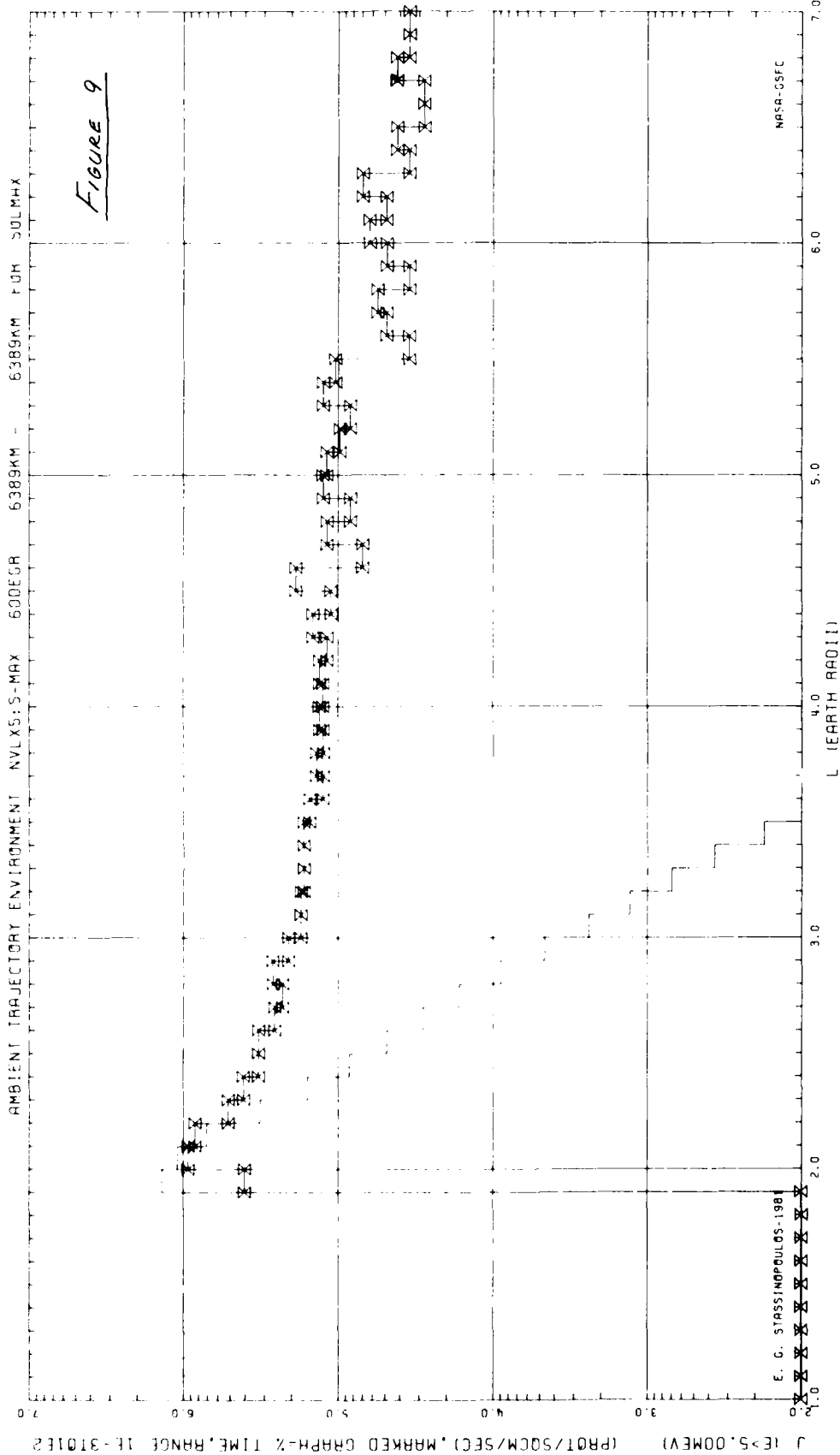
(PLOT/SOCM/SECI) MARKED GRAPH: TIME RANGE 11-3101E2 (E>5.00MEV)

AMBIENT TRAJECTORY ENVIRONMENT NV 4345 MAX 600EGR 5186KM 5186KM FOR 50R MVA

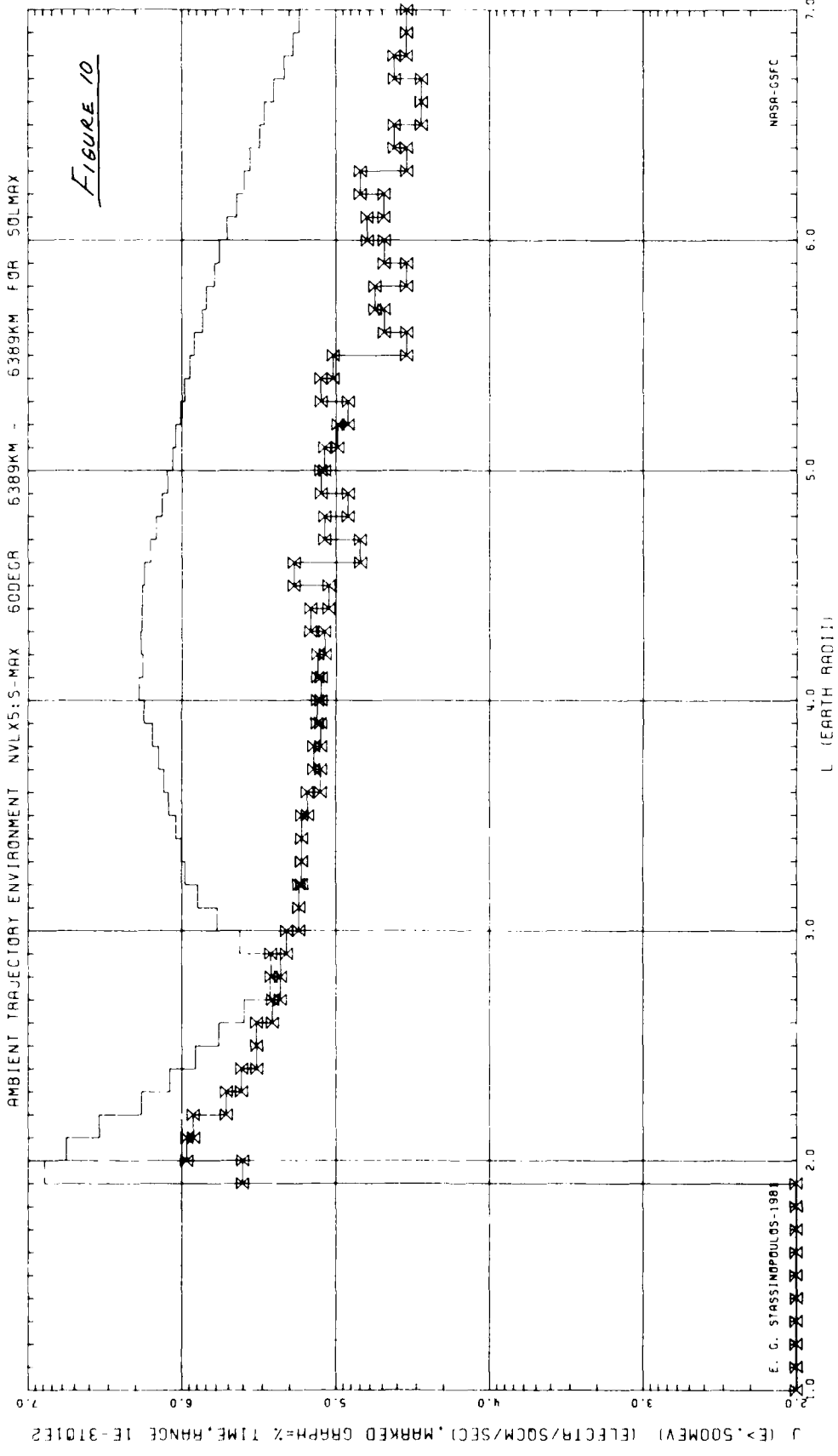
Figure 8







J (>5.00MEV) (PROT/SQCM/SEC), MARKED GRAPH-% TIME, RANGE 1E-3101E2



AMBIENT TRAJECTORY ENVIRONMENT NVLX6:S-MAX 60DEC8 10371KM - 10371KM FOR SOLMAX

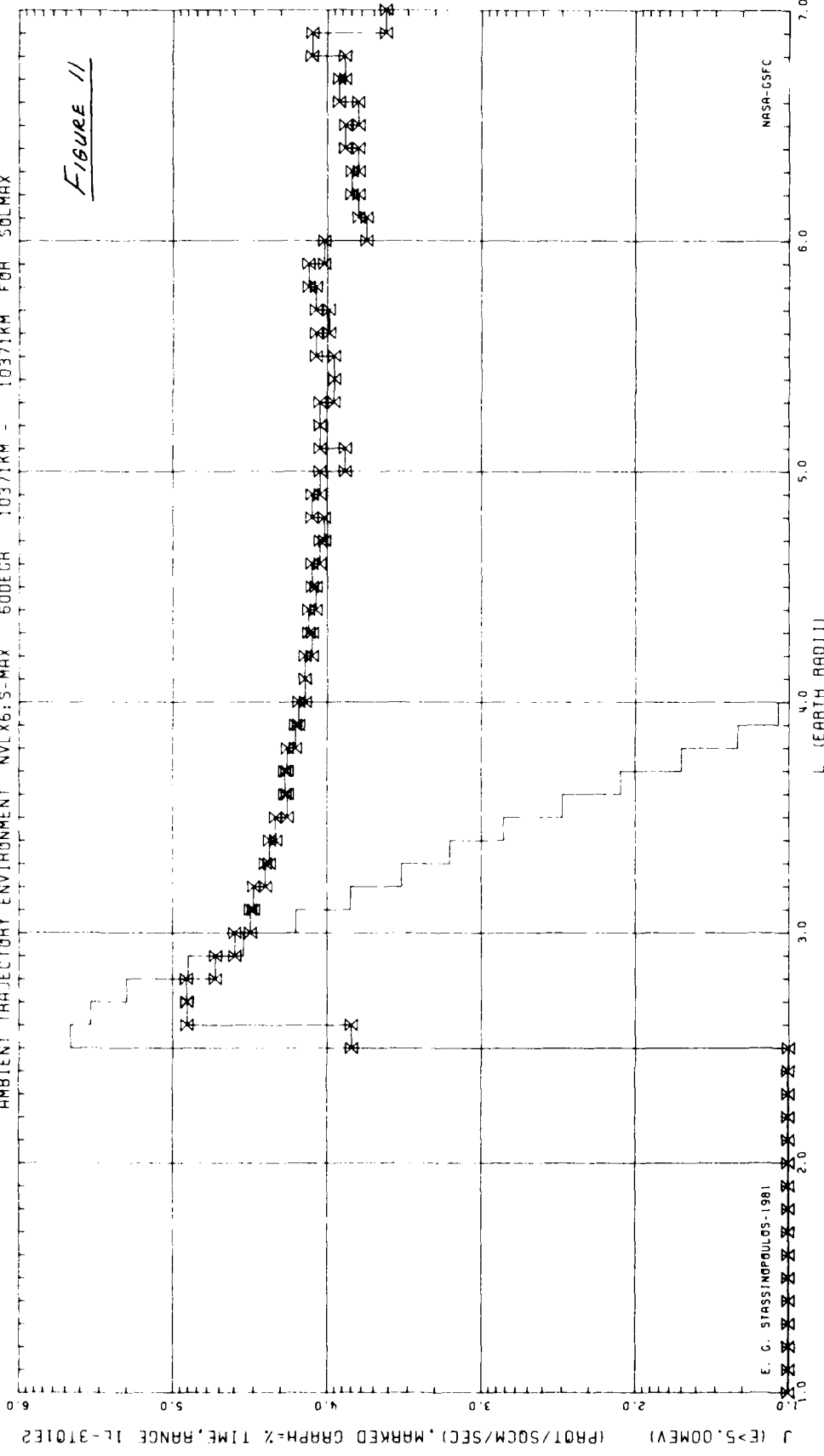
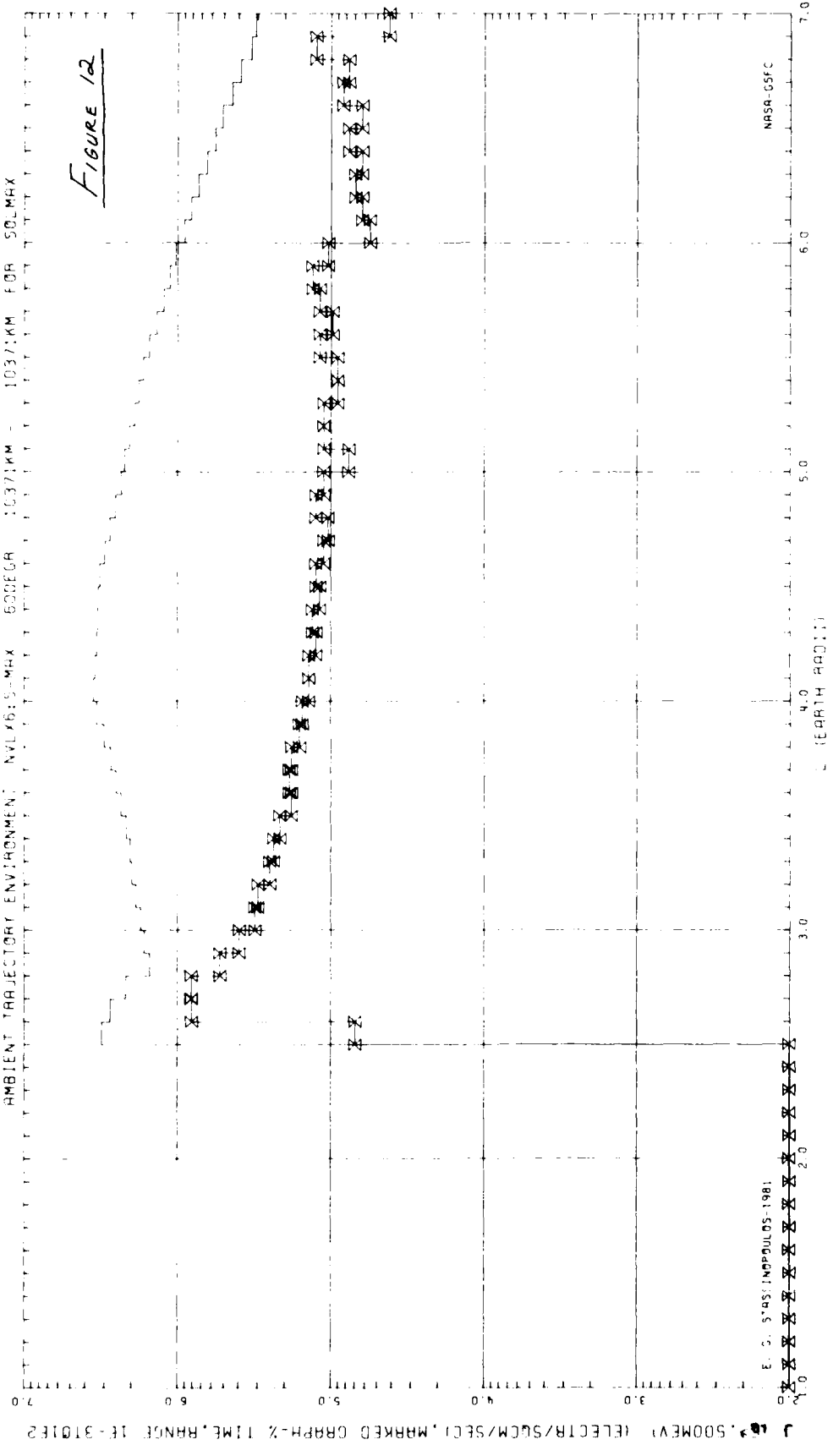


Figure 11

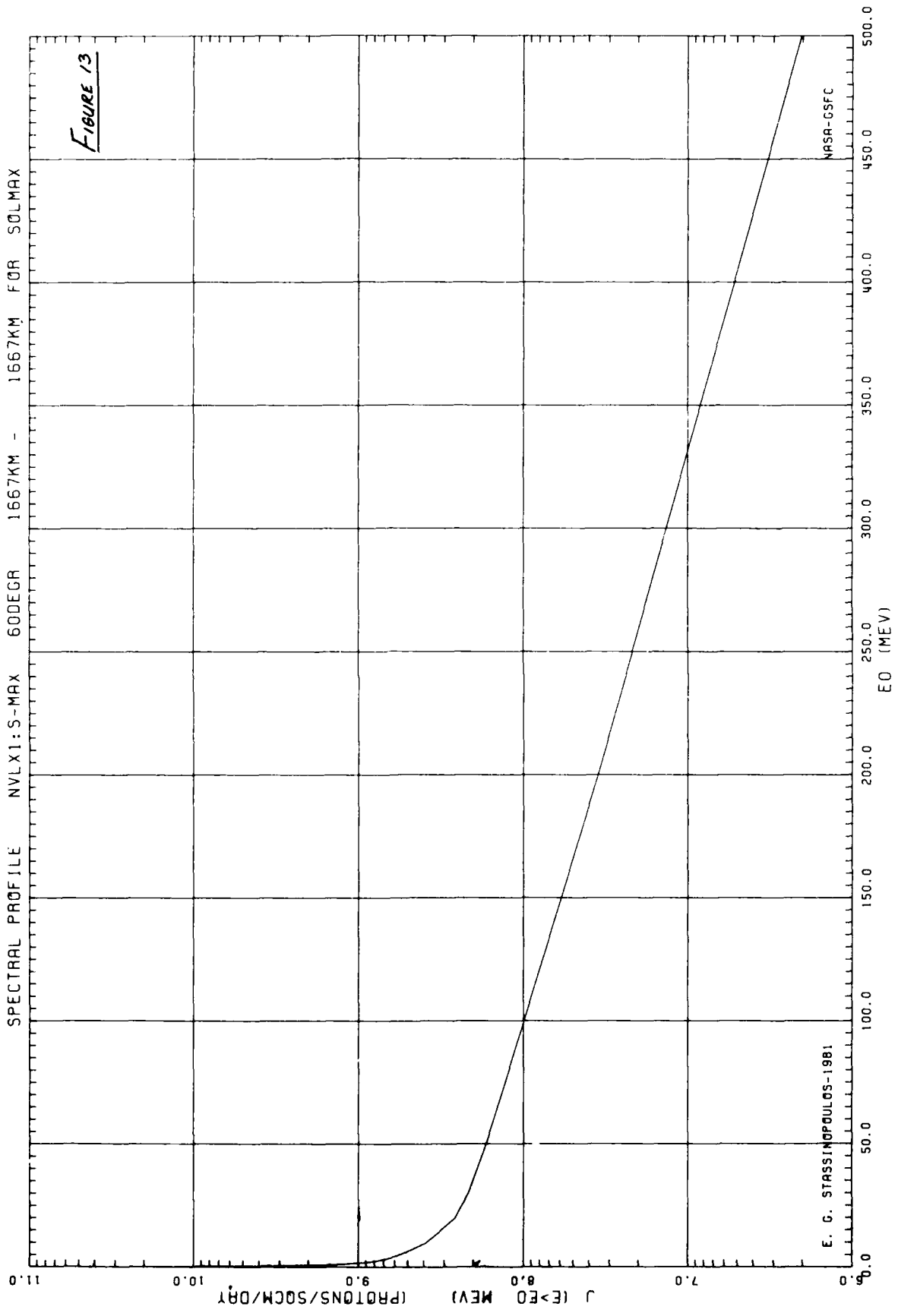
AMBIENT TRAJECTORY ENVIRONMENT: N41X61S-MAX 600EGR 10371KM 10371KM FOR SOL-MAX

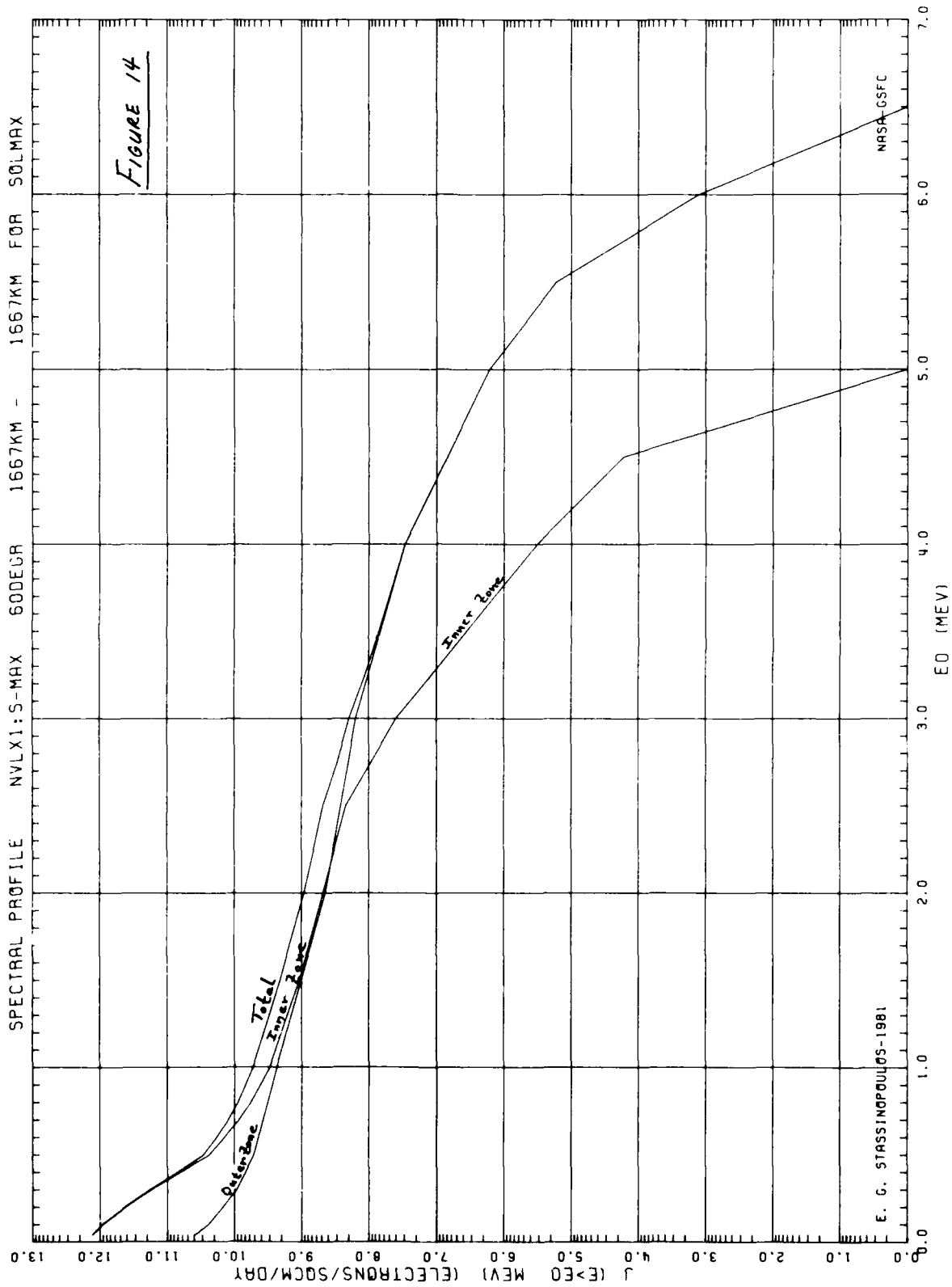
FIGURE 12



E. S. 5'851 INPAULOS-1981

NASA-OSFC





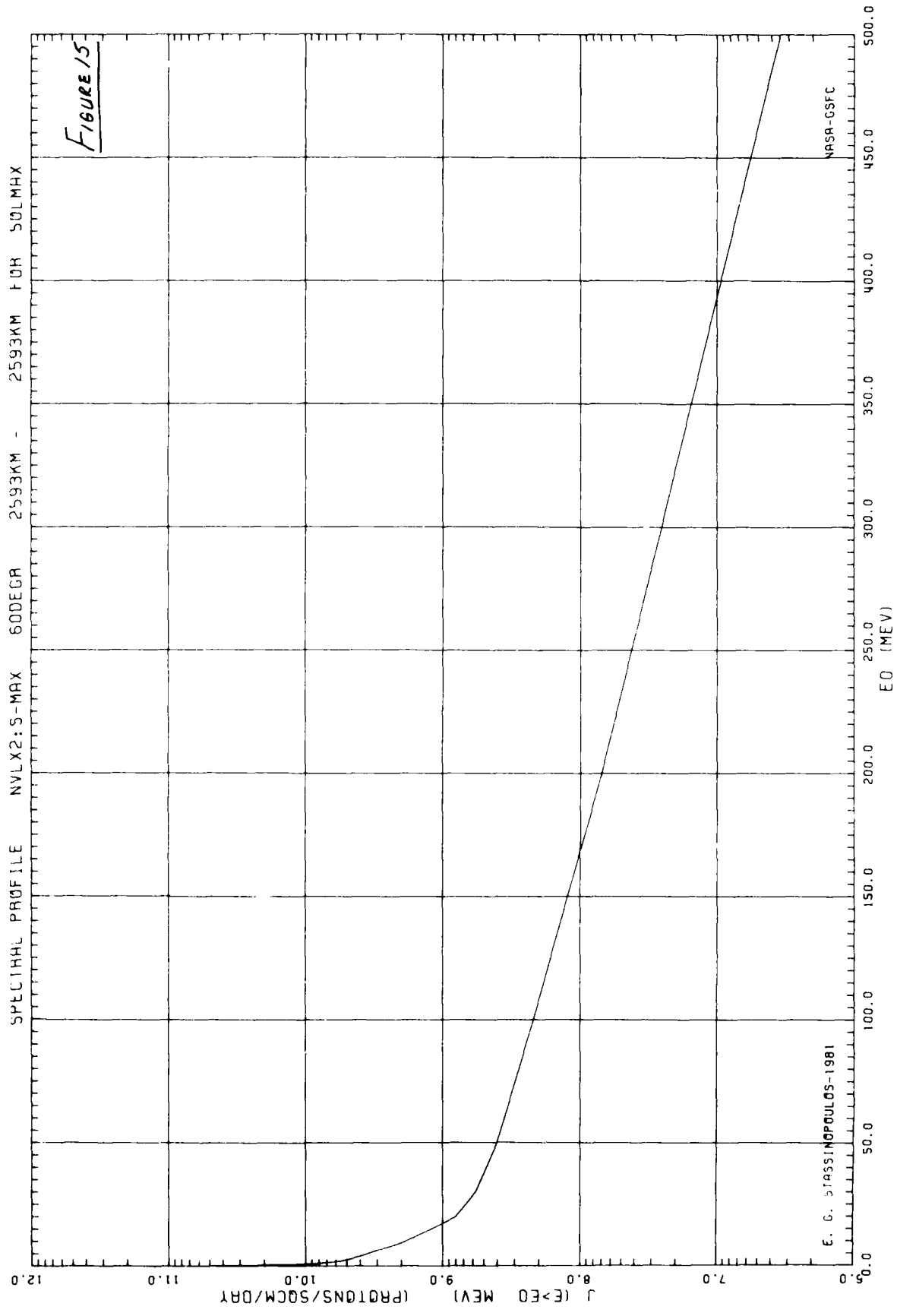


FIGURE 15

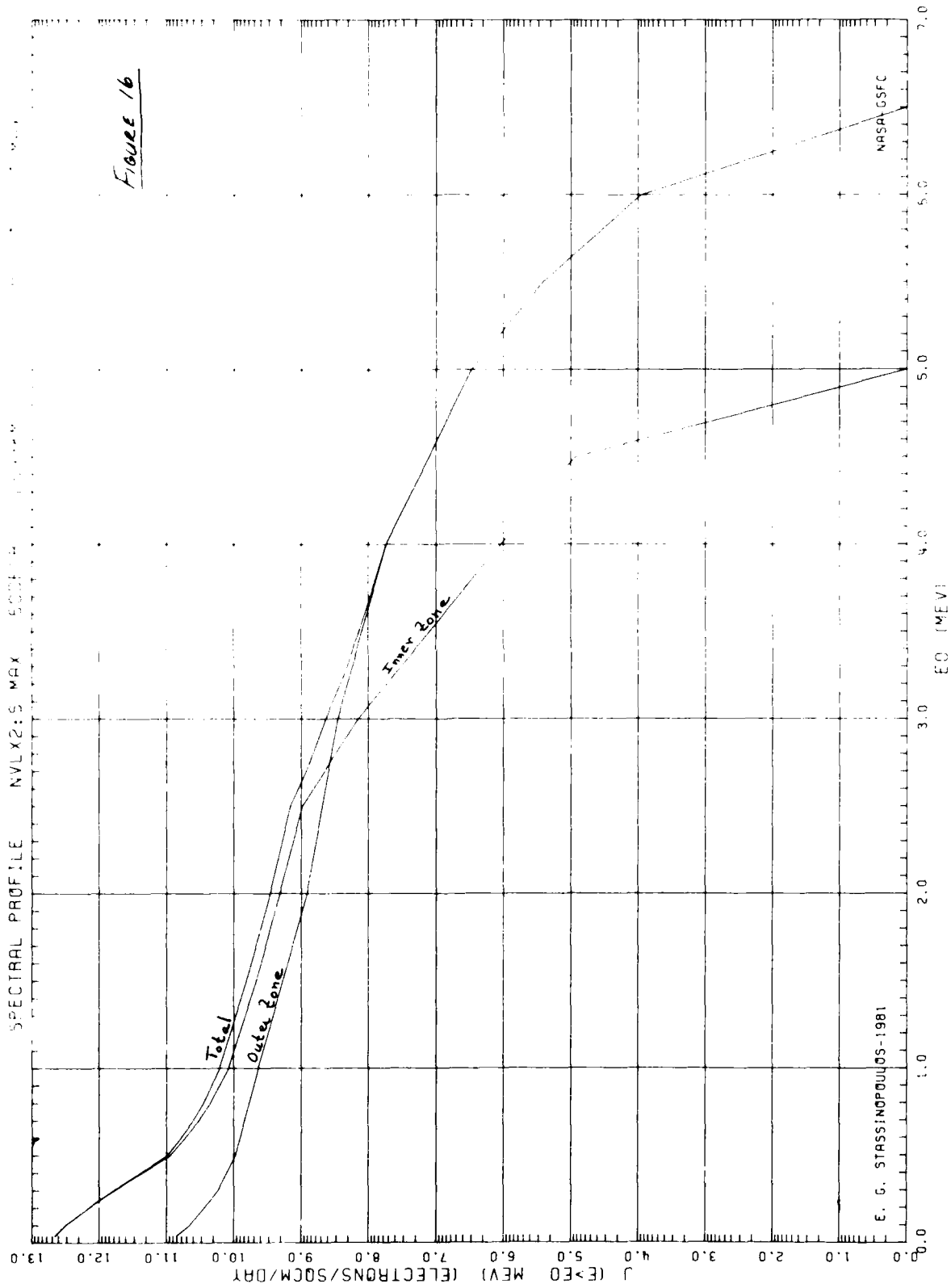
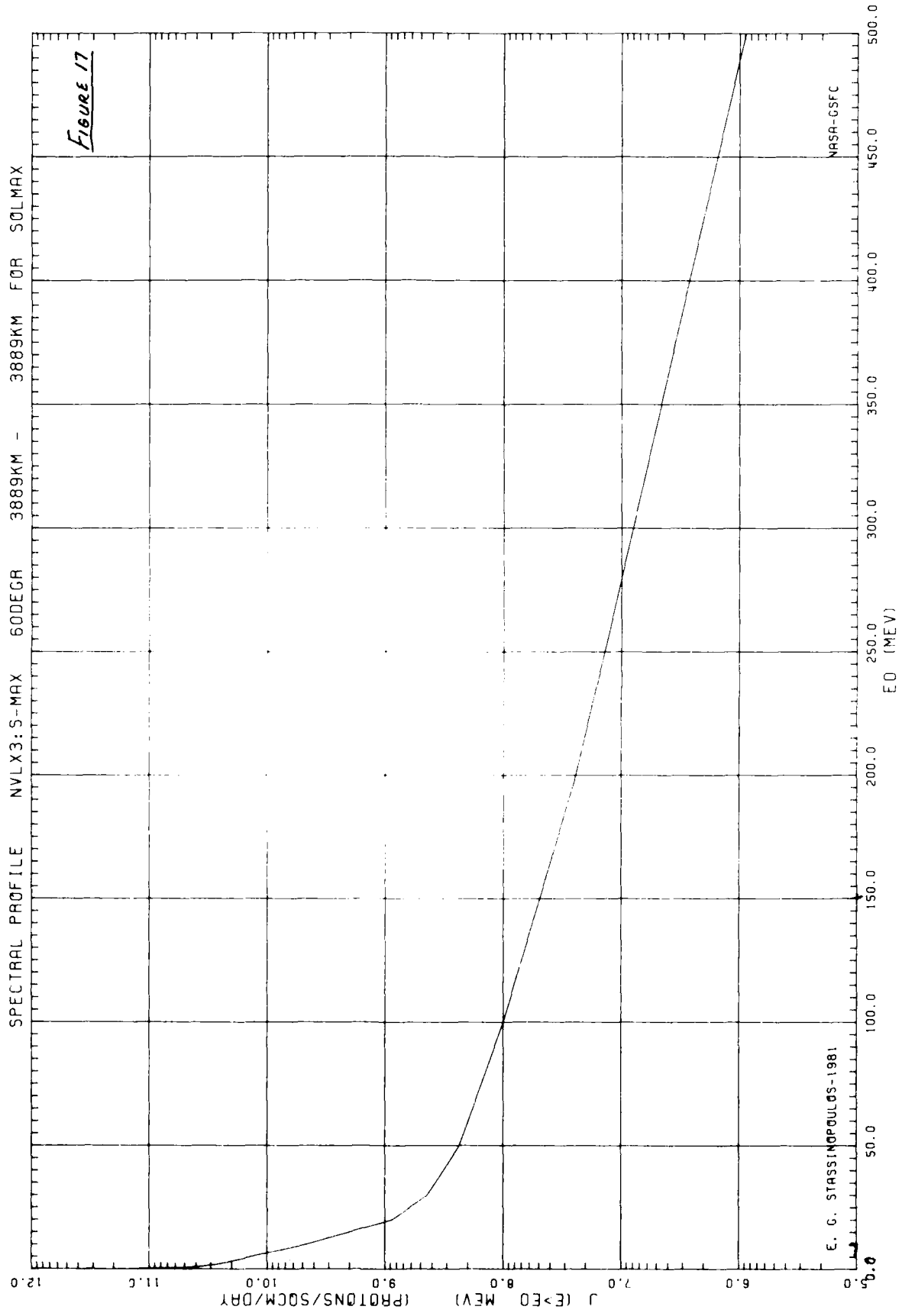
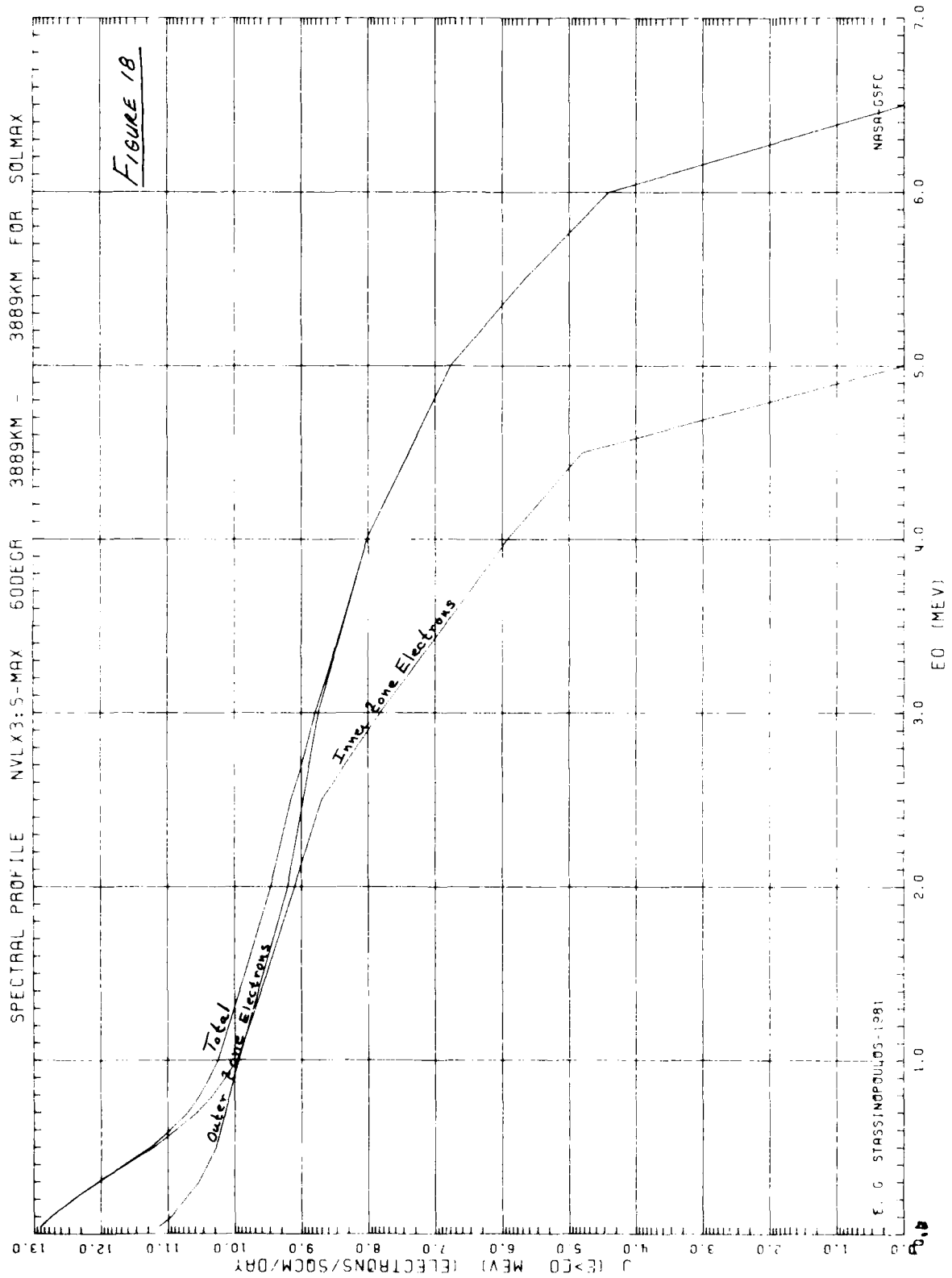


Figure 16







AD-A141 849

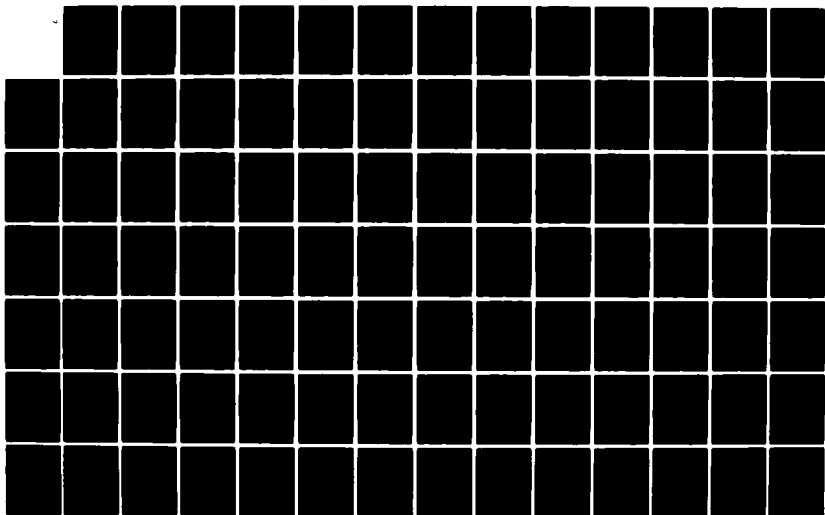
ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR  
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE  
ADMINISTRATION GREENBELT MD GO. . E G STASSINOPOULOS  
NOV 81 NASA-GSFC-X-601-81-28

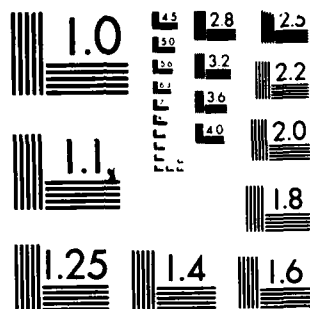
2/5

UNCLASSIFIED

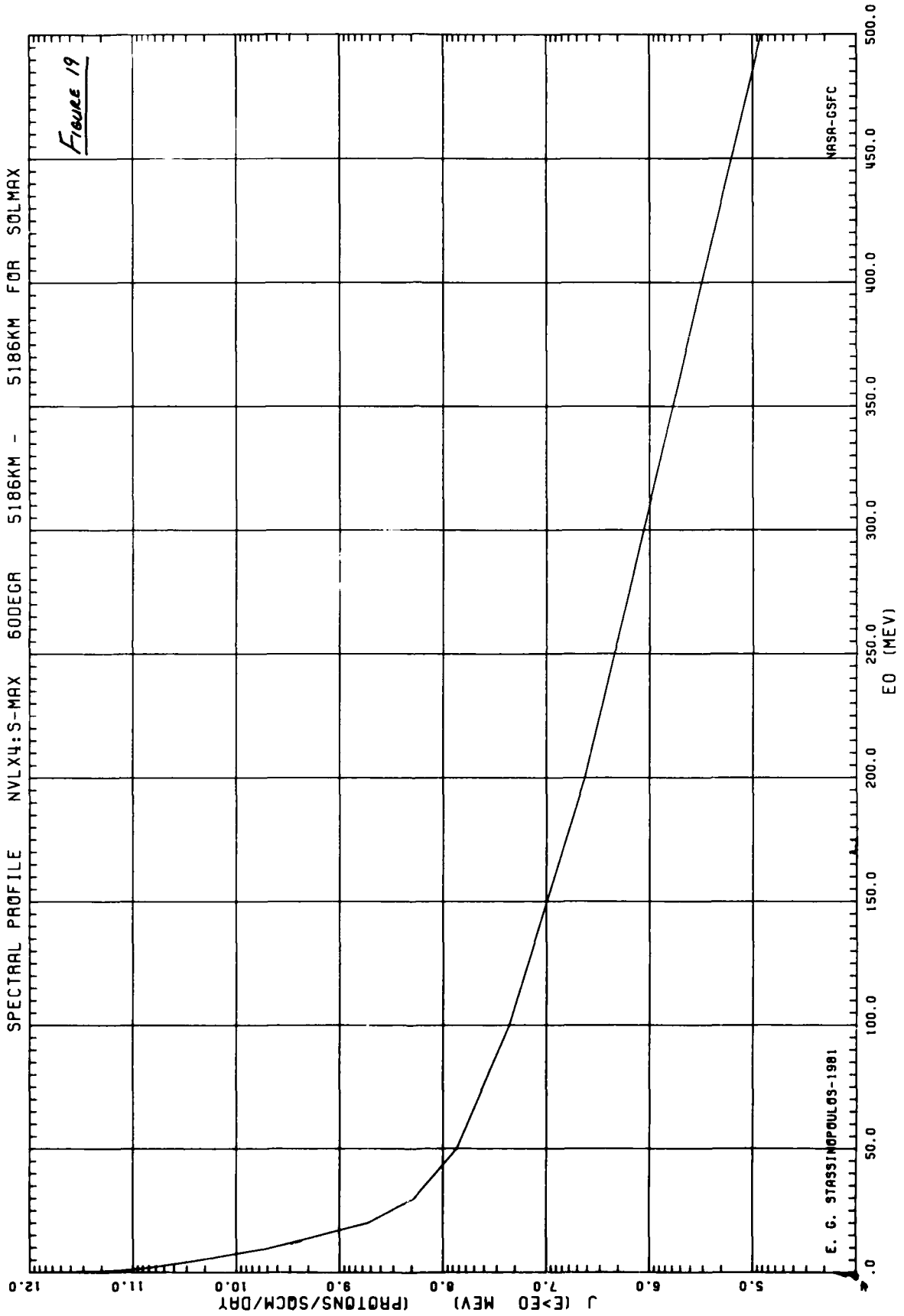
F/G 22/3

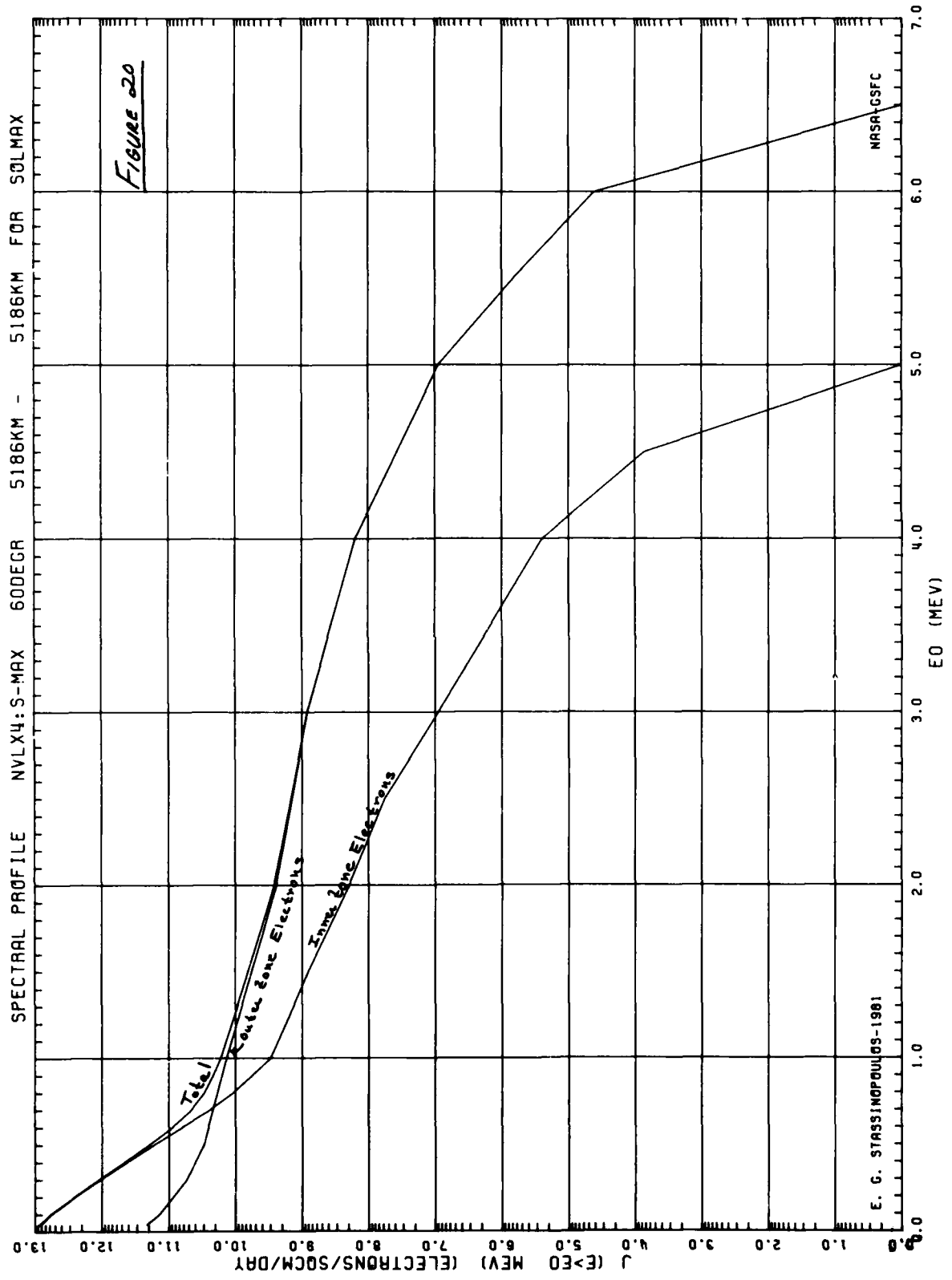
NL

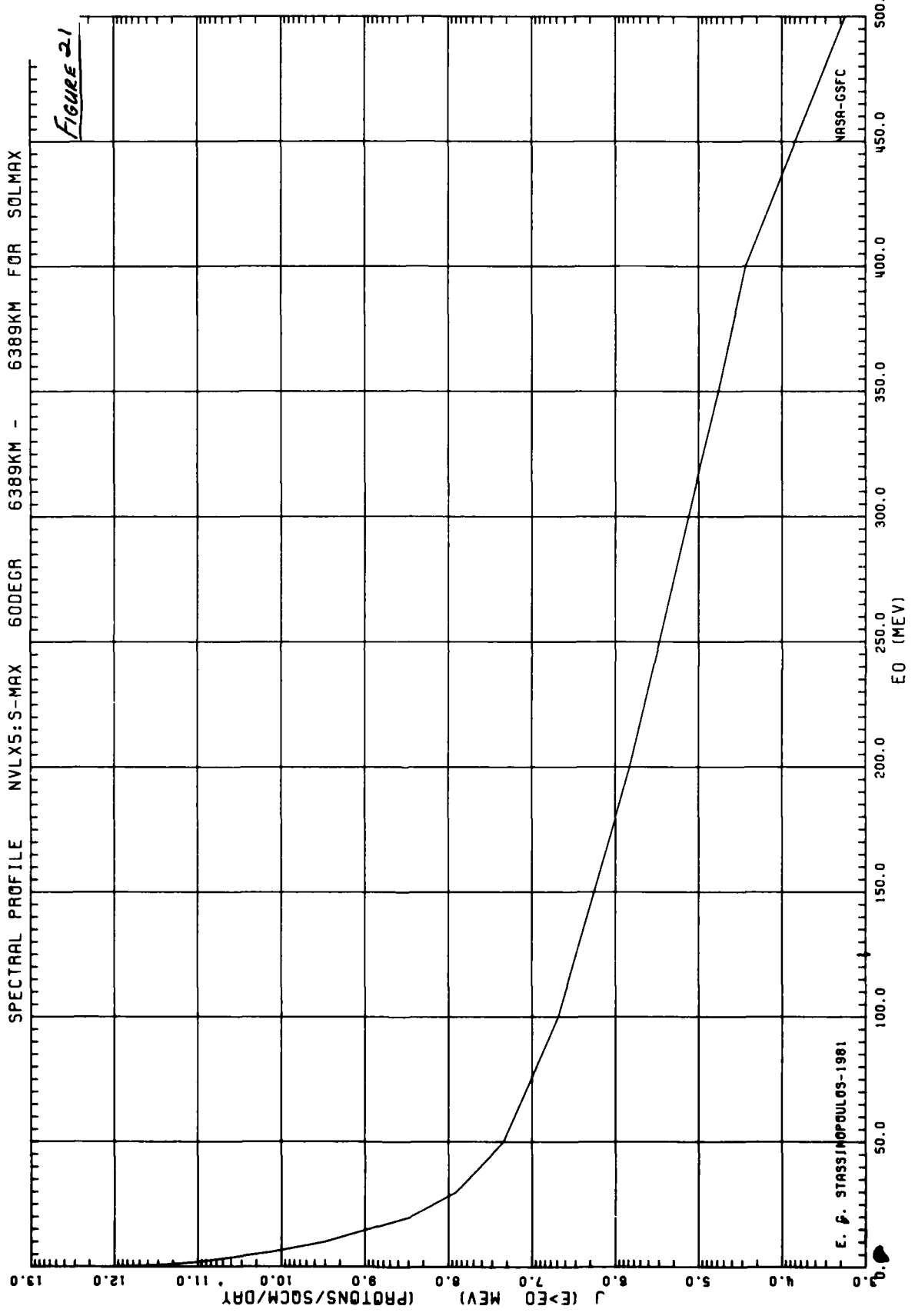


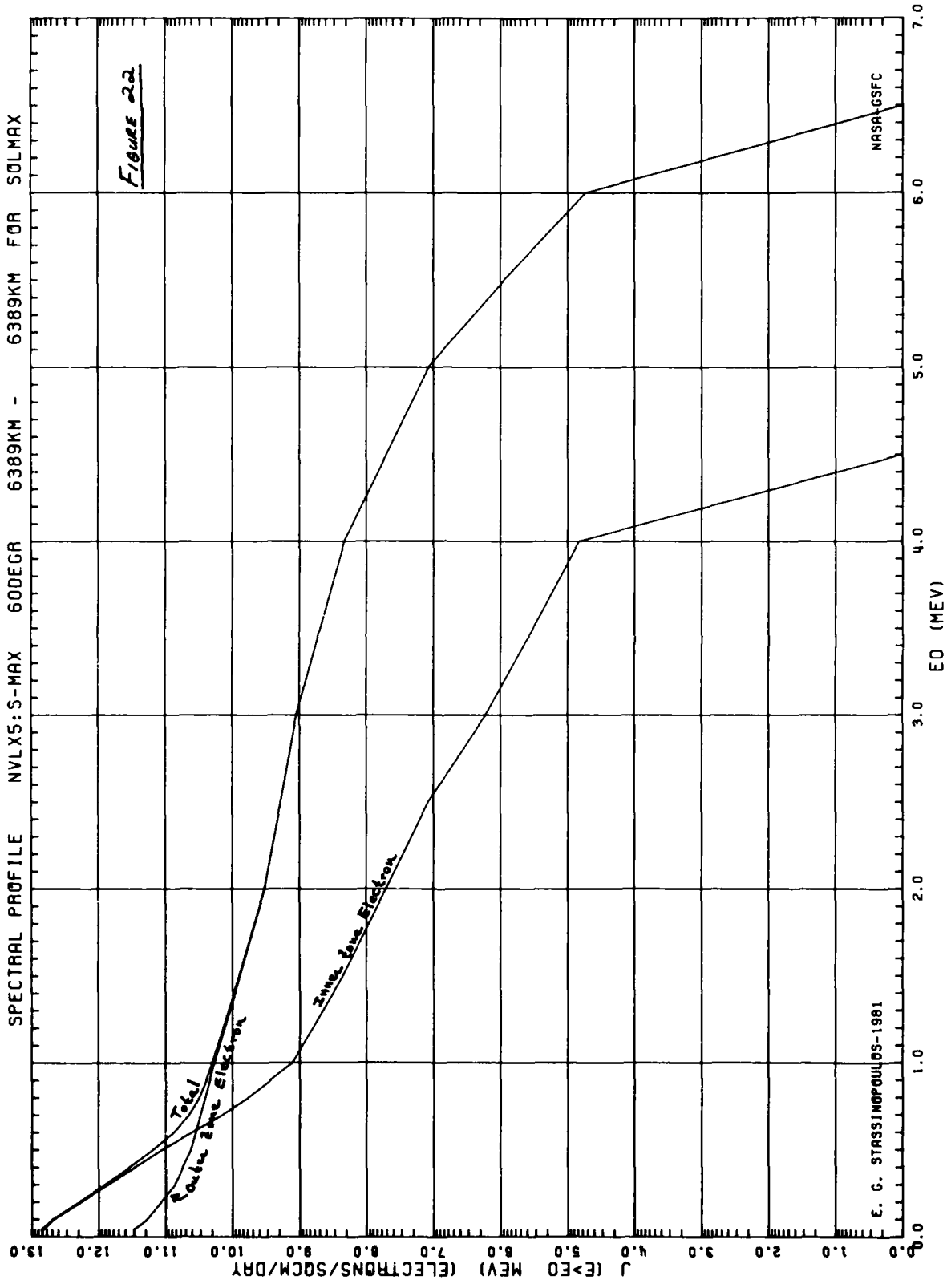


MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS 1963 A

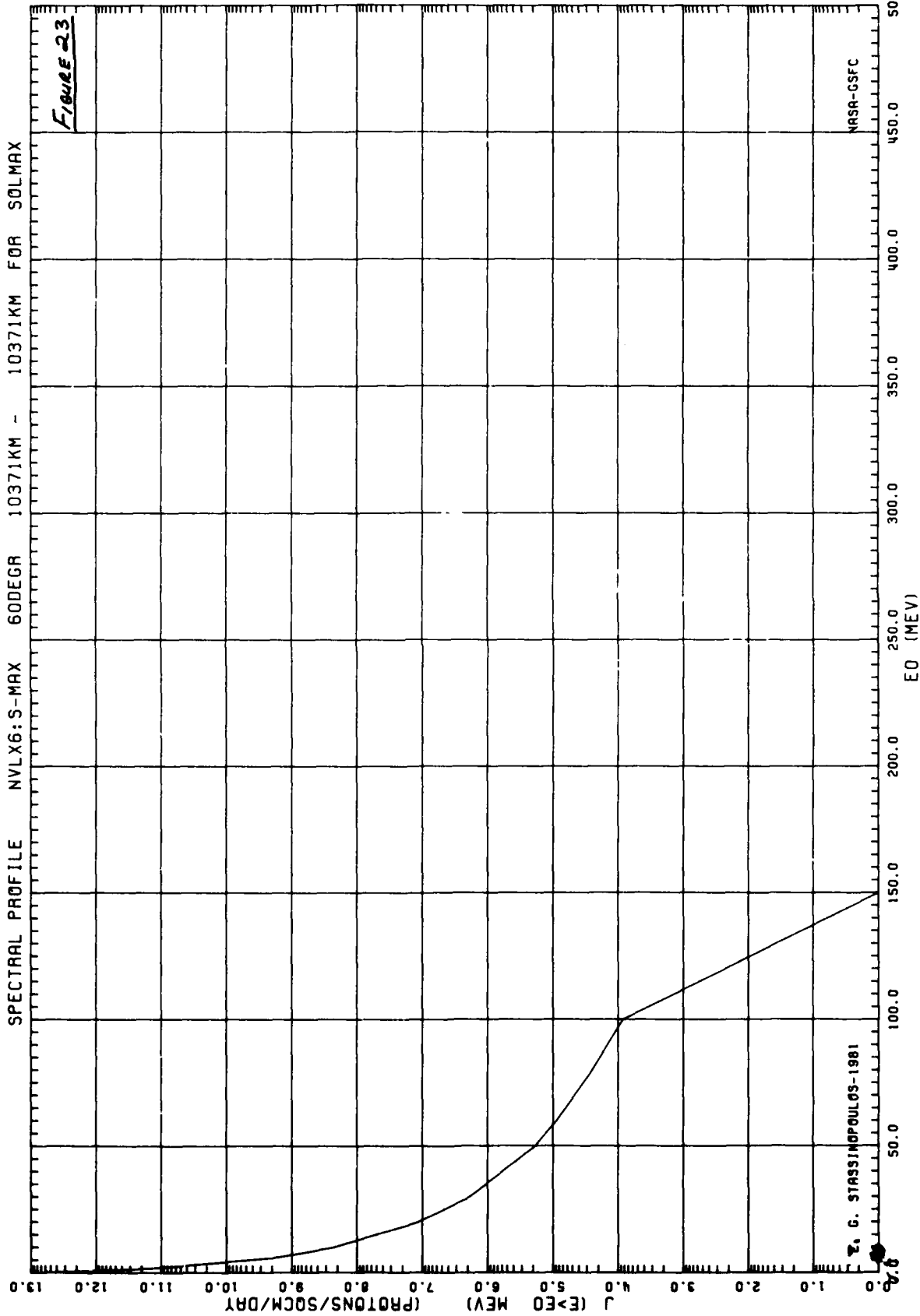


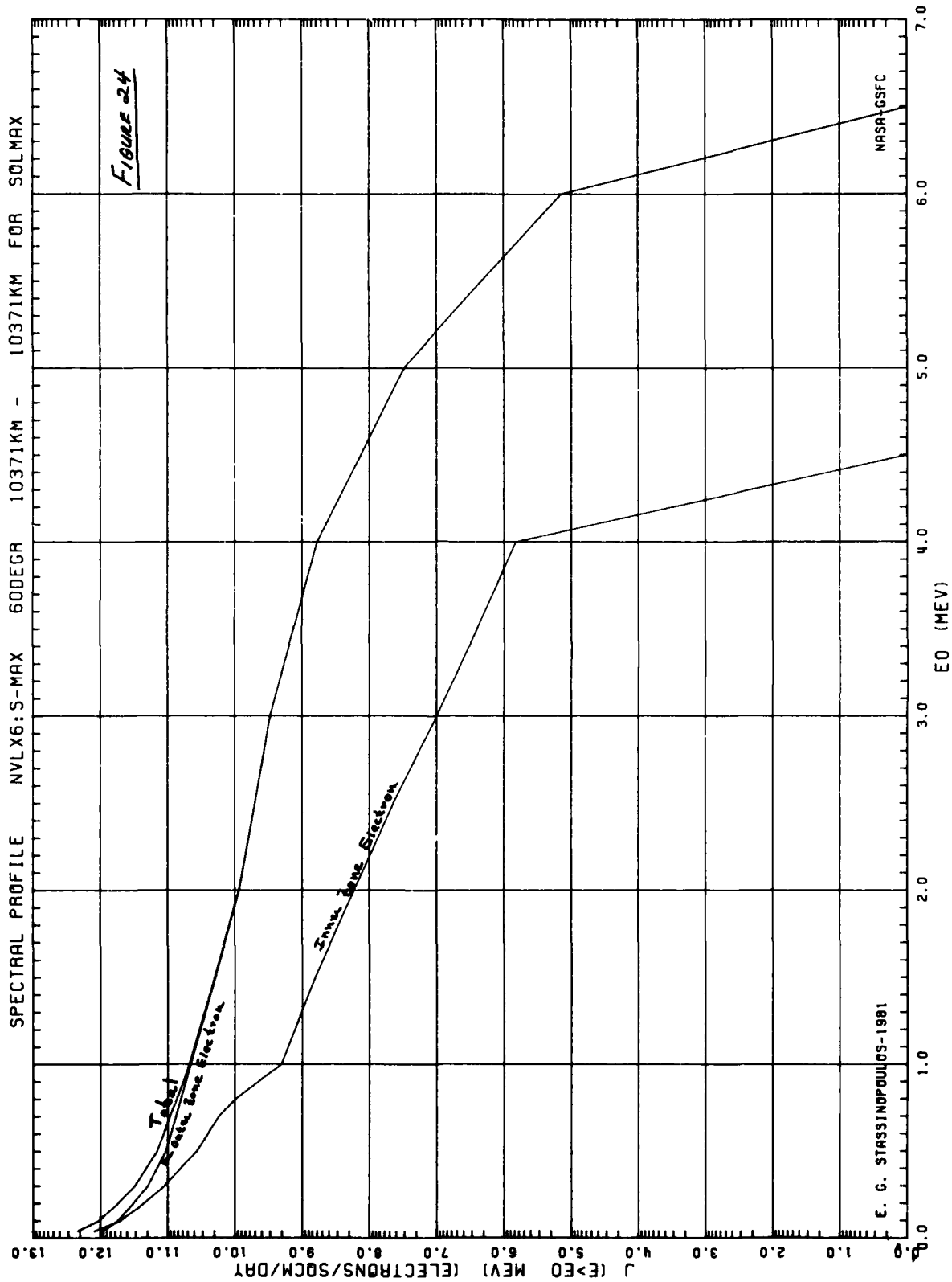


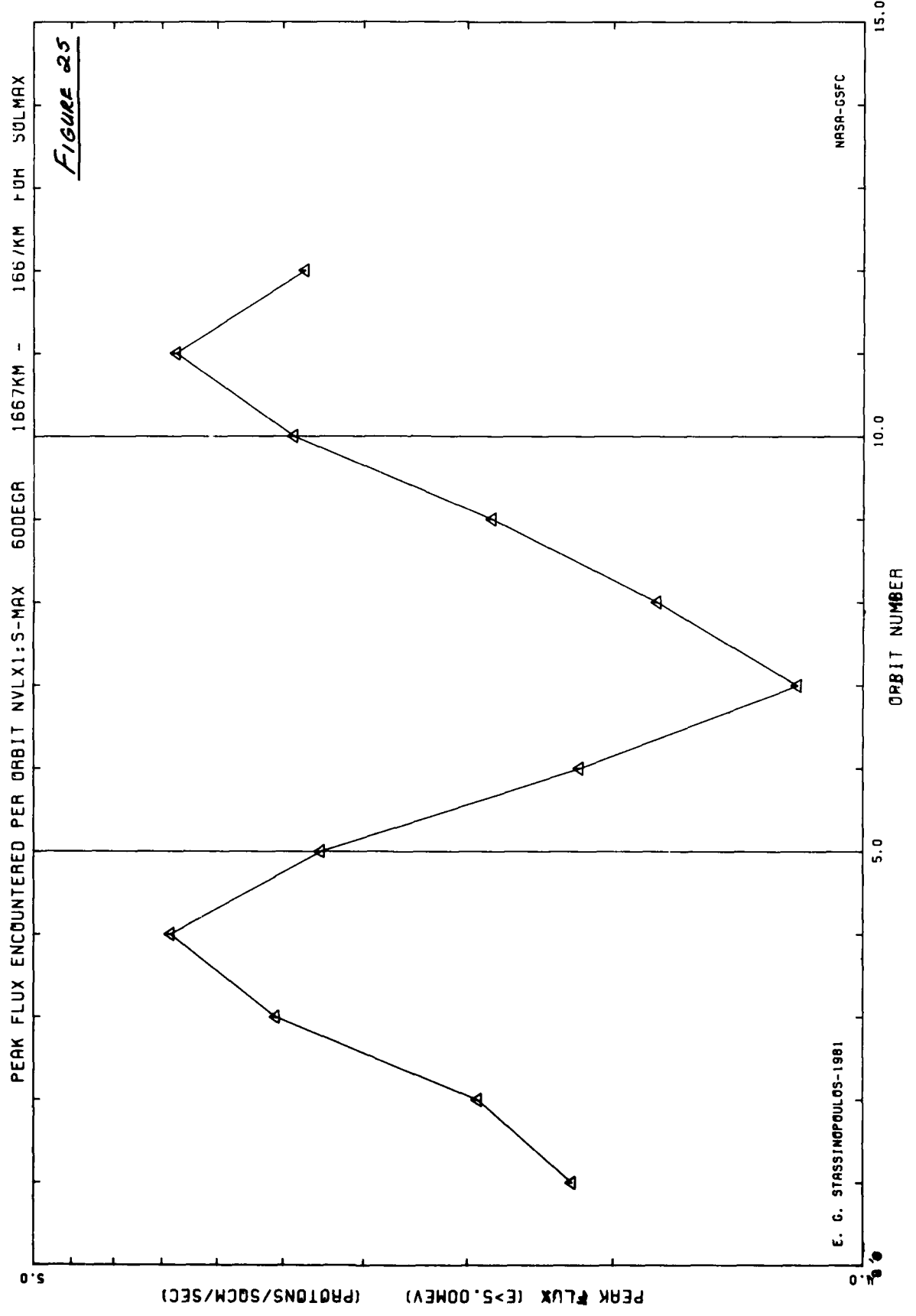


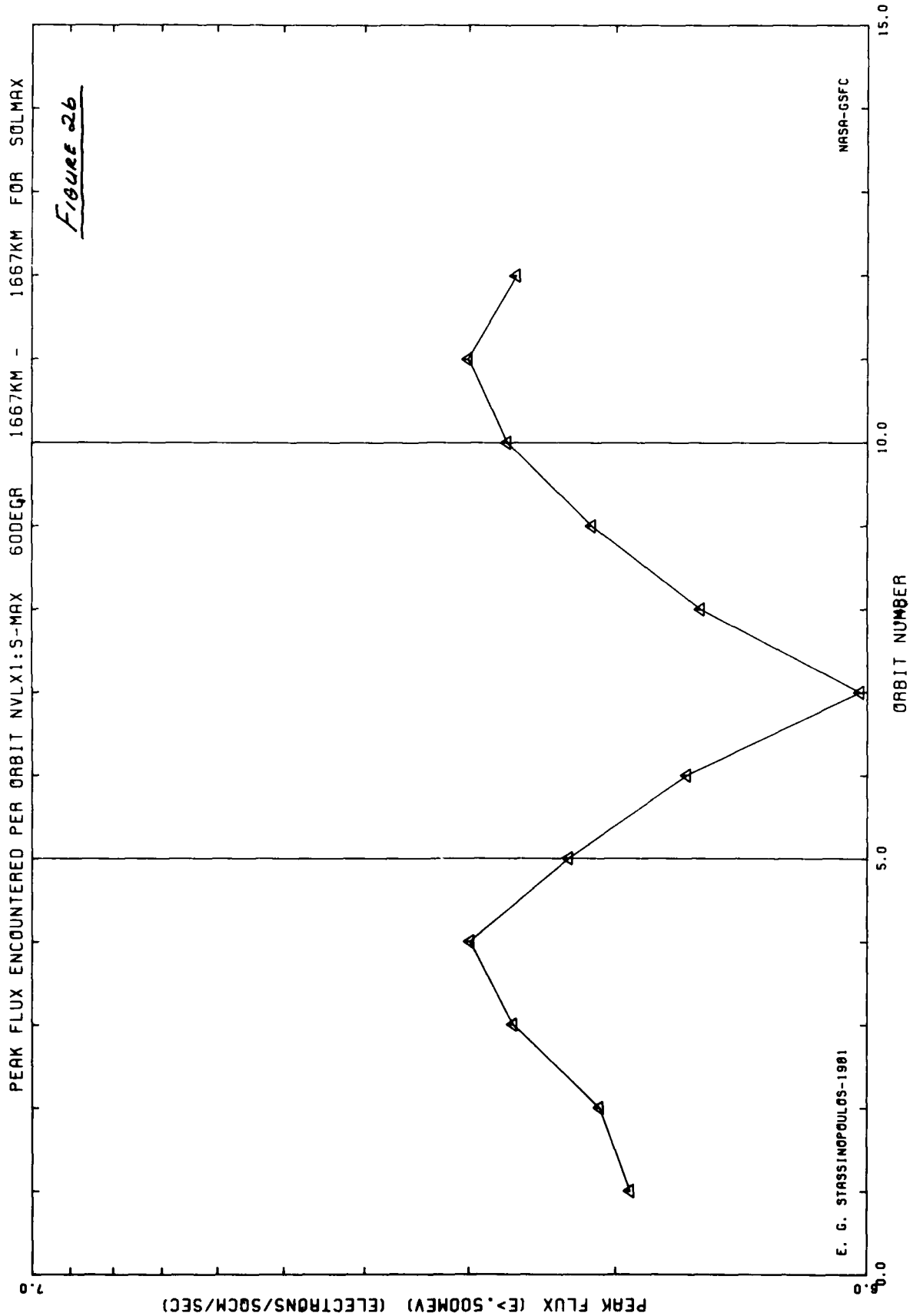






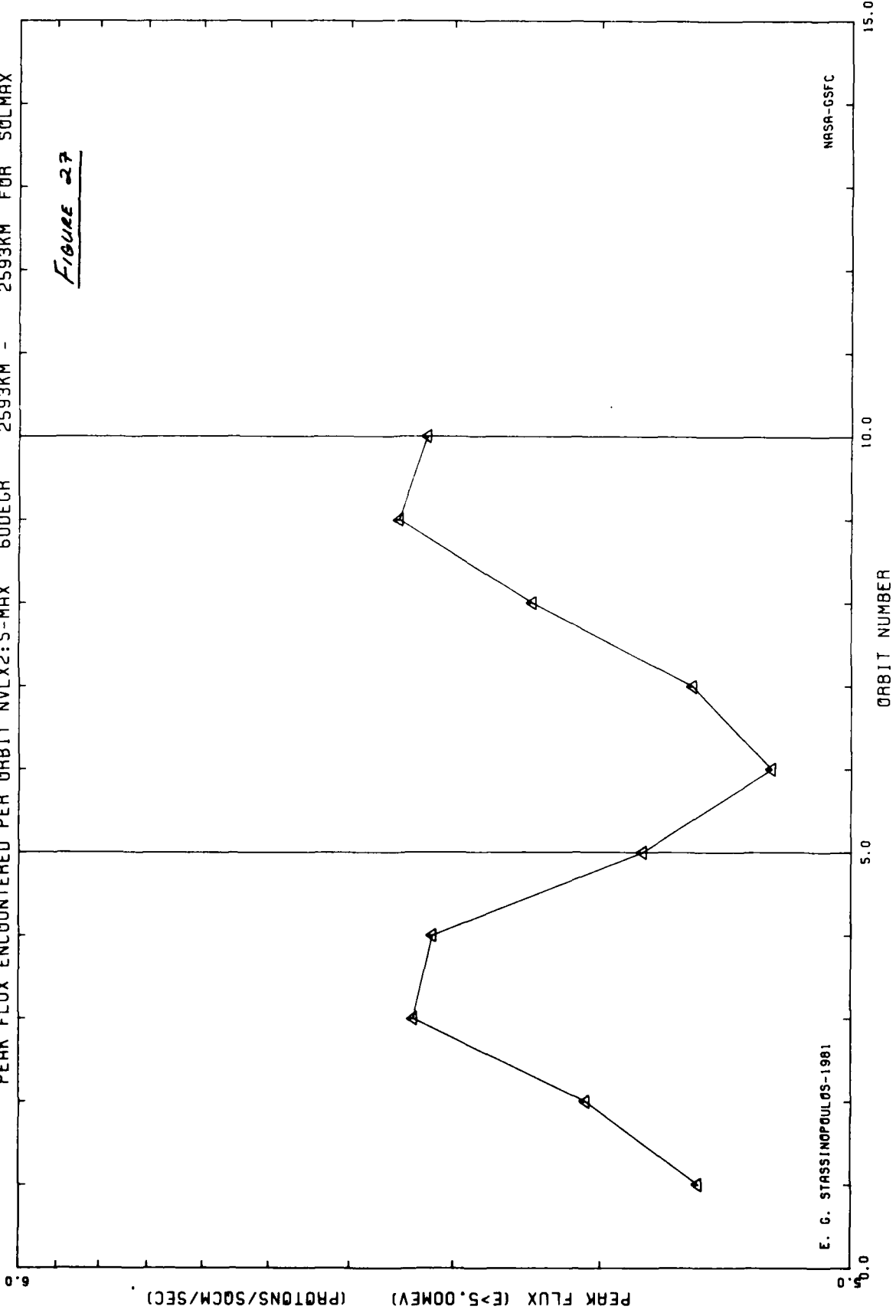






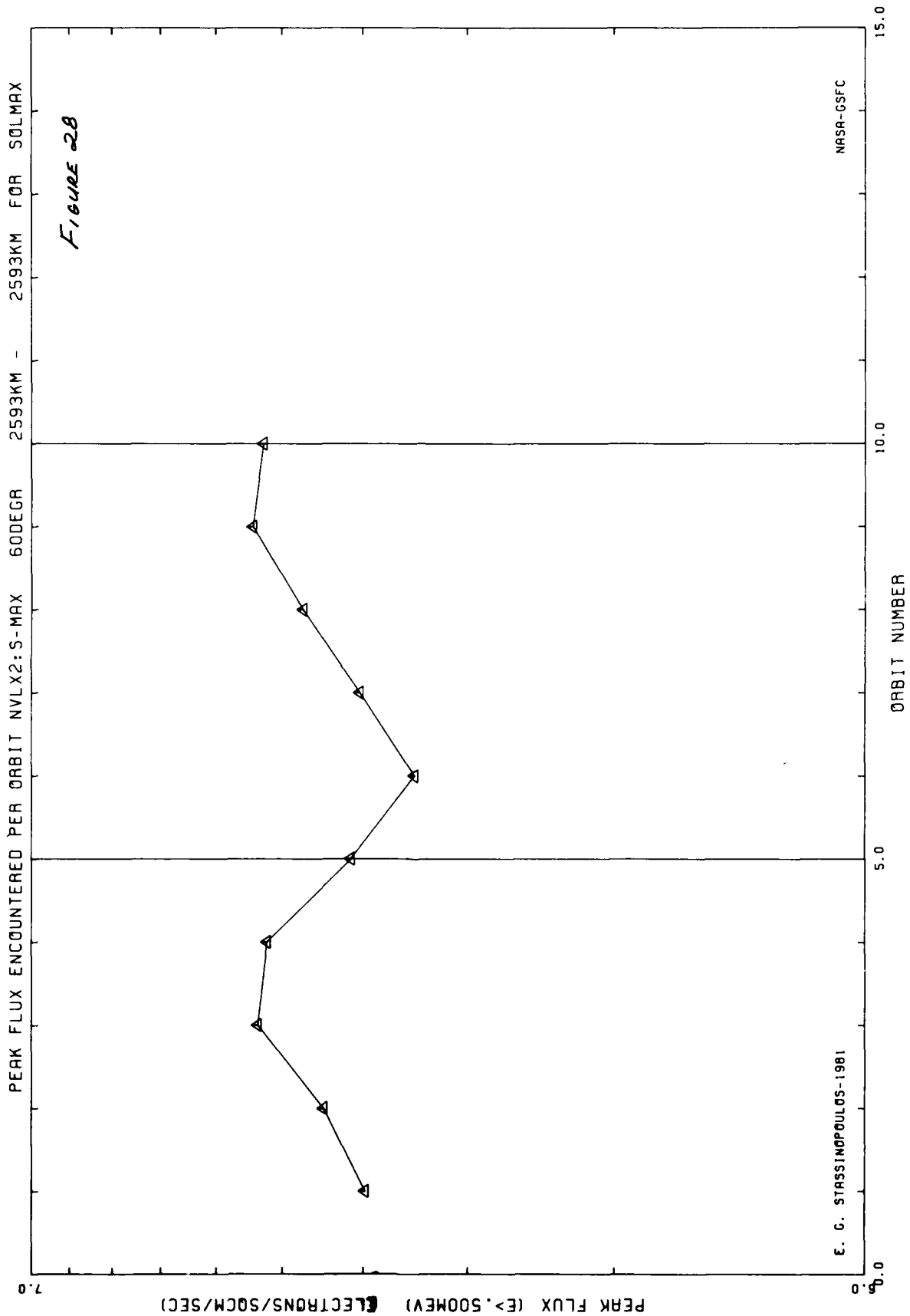
PEAK FLUX ENCOUNTERED PER ORBIT NVLX2: S-MAX 60DEGR 2593KM - 2593KM FOR SOLMAX

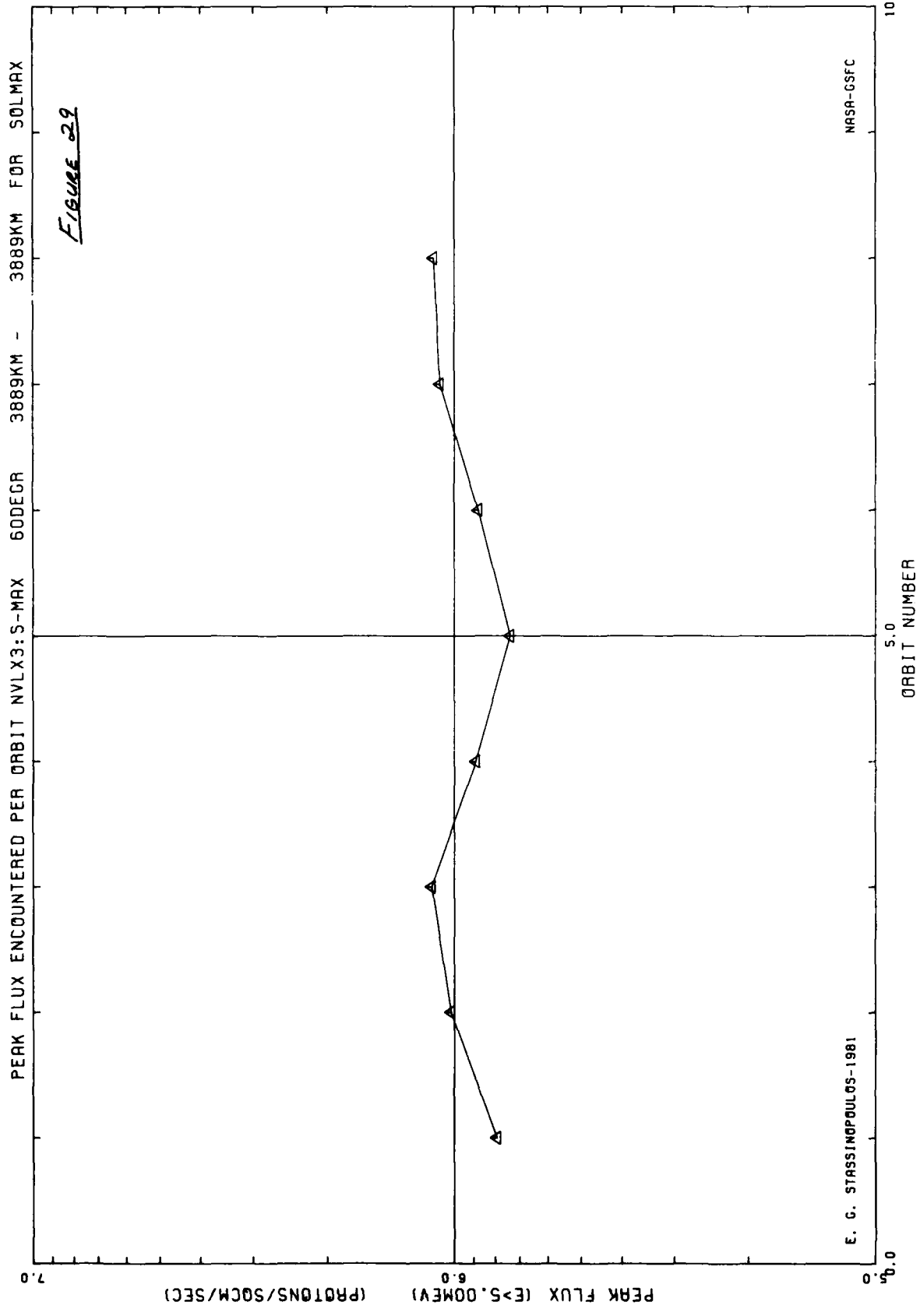
Figure 27



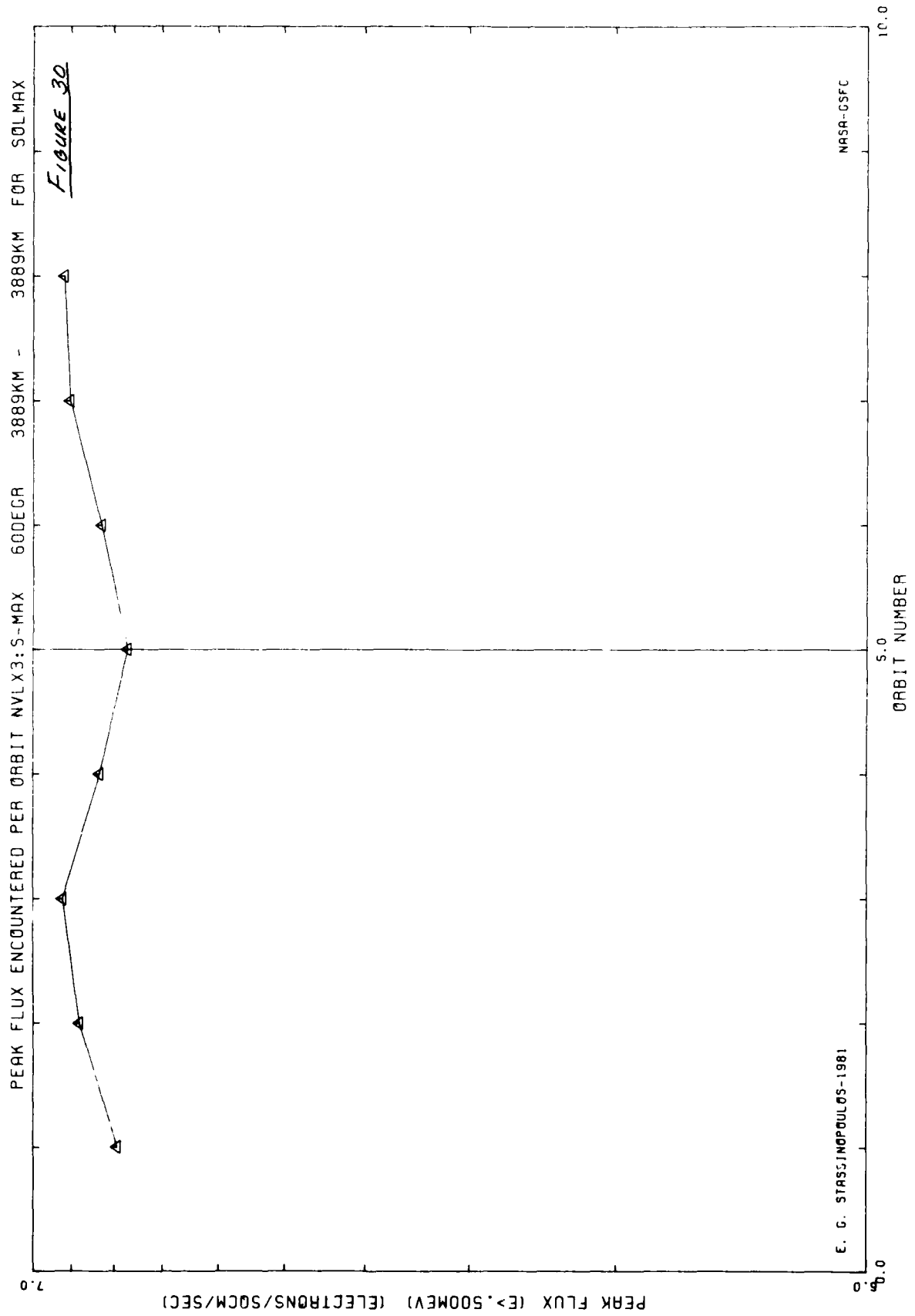
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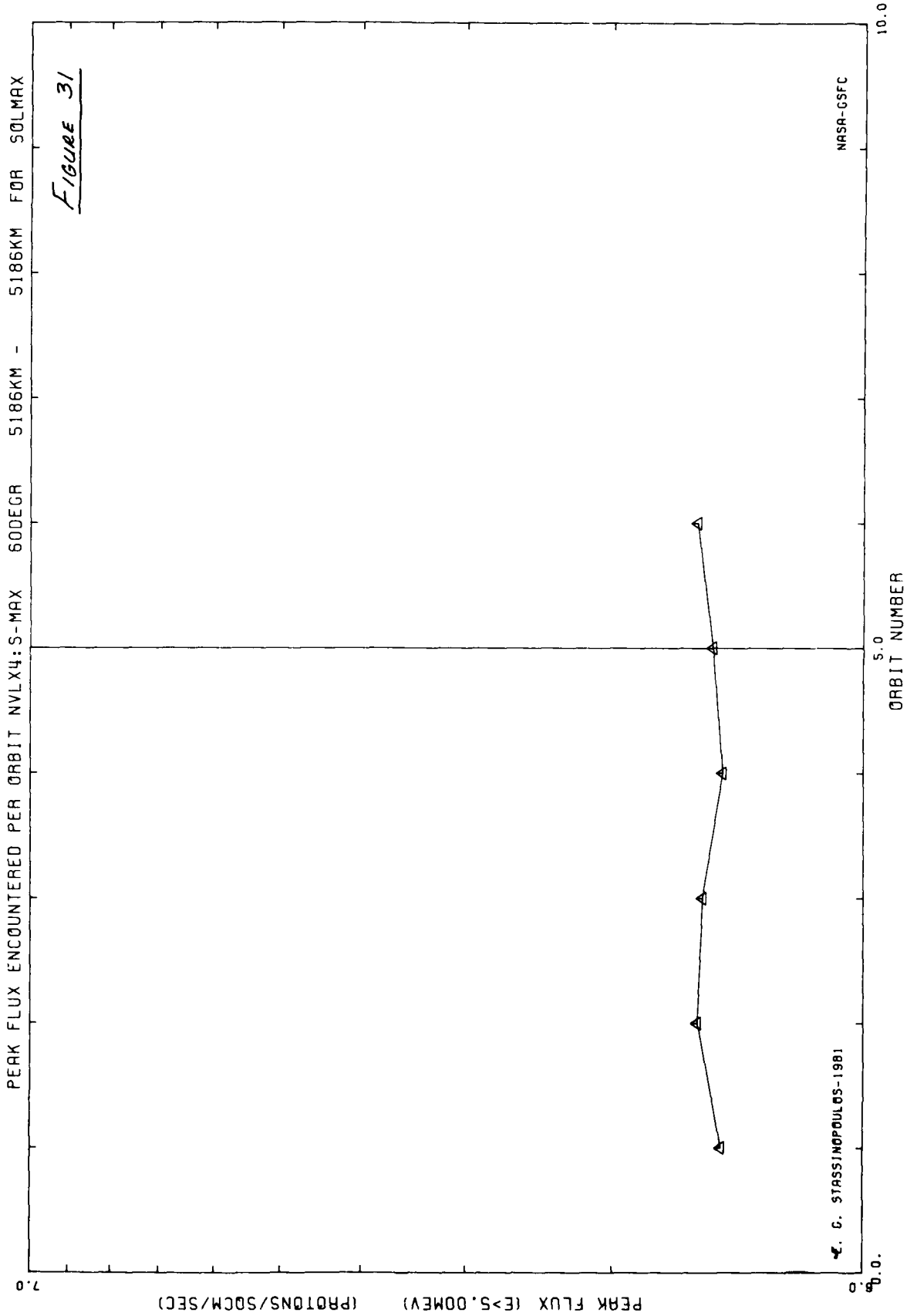




E. C. STASSINGPOULOS-1981

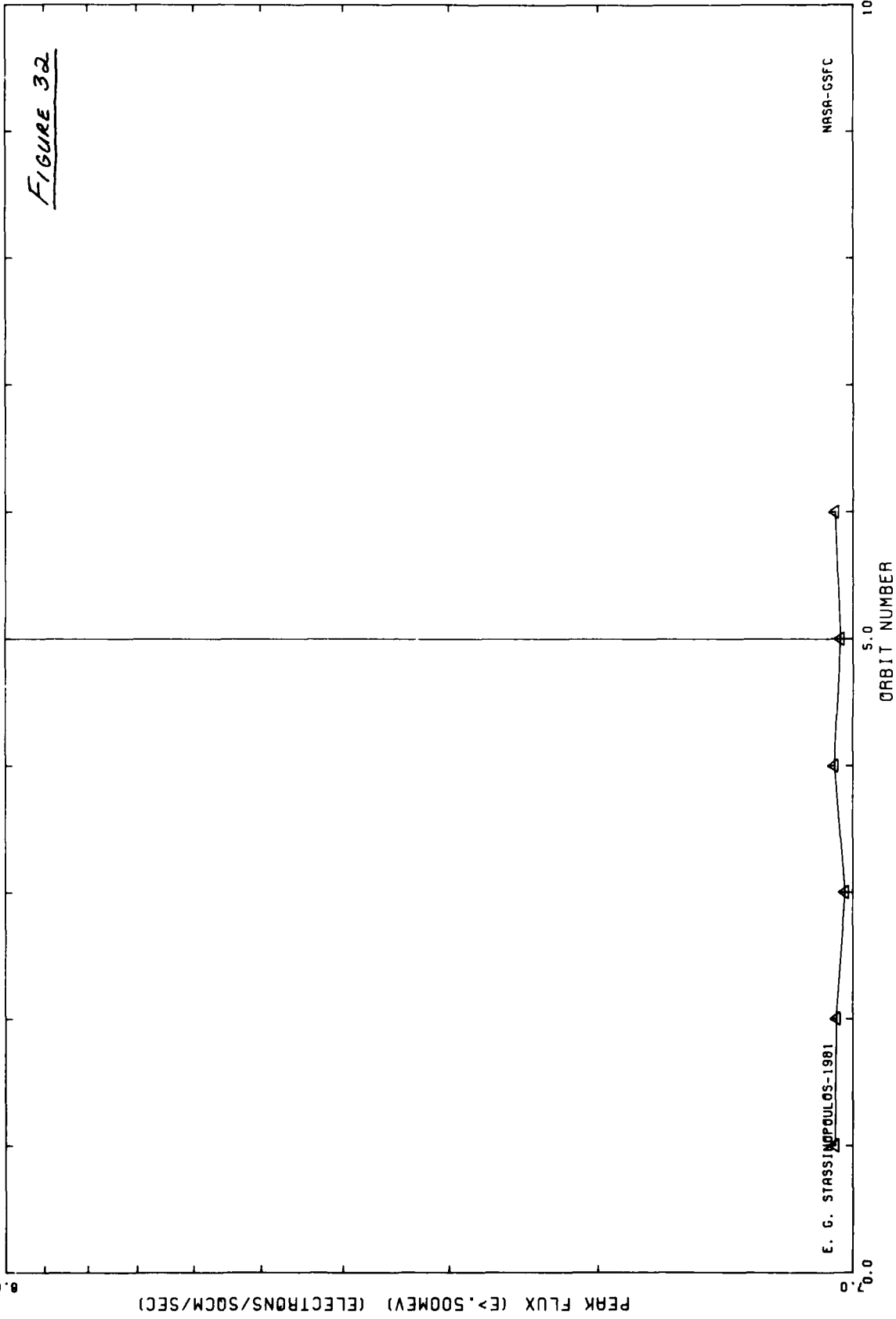






PEAK FLUX ENCOUNTERED PER ORBIT NVLX4: S-MAX 60DEGR 5186KM - 5186KM FOR SOLMAX

Figure 32

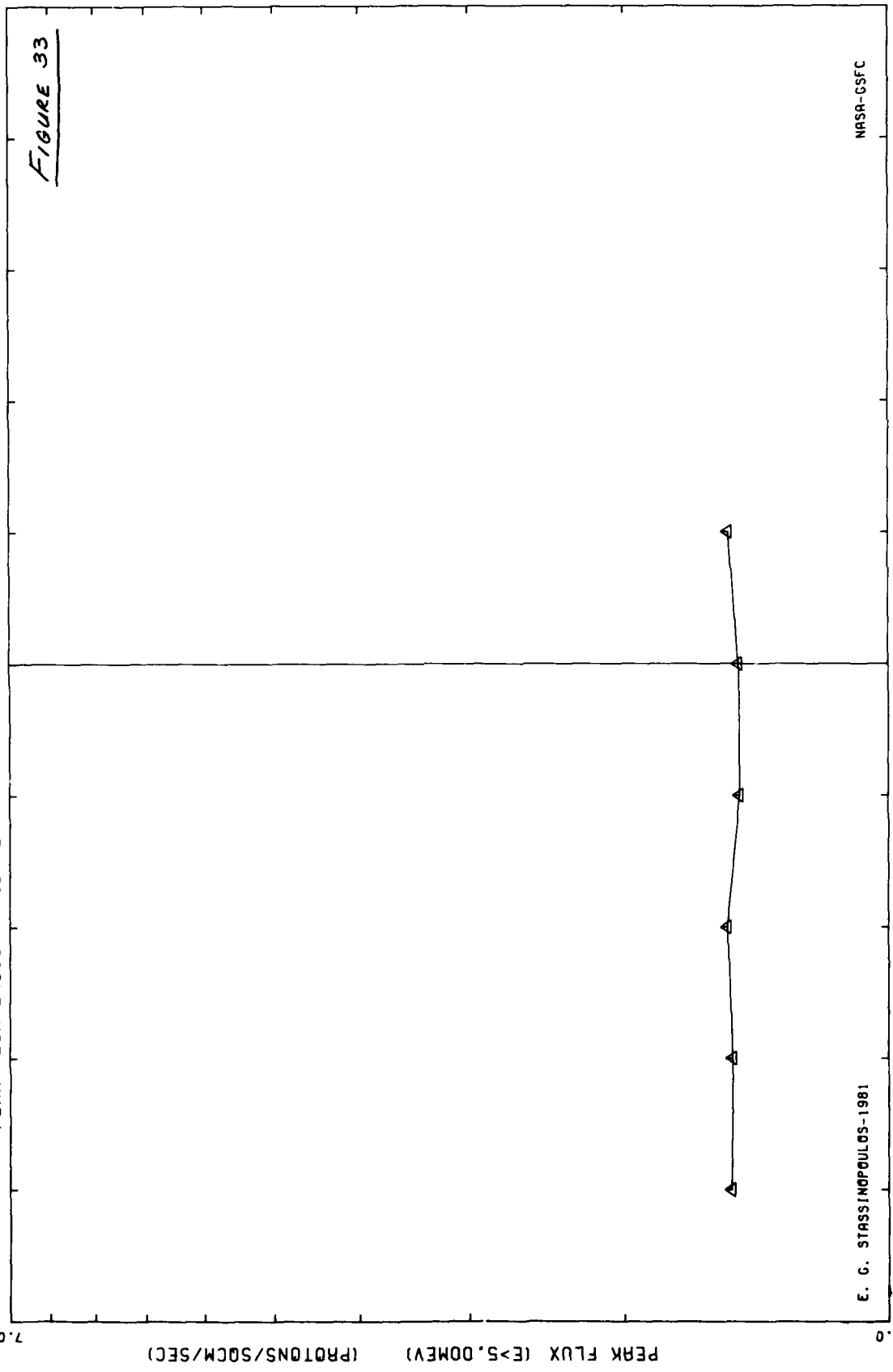


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PEAK FLUX ENCOUNTERED PER ORBIT NVLX5:S-MAX 60DECR 6389KM - 6389KM FOR SOLMAX

FIGURE 33



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10.0

5.0

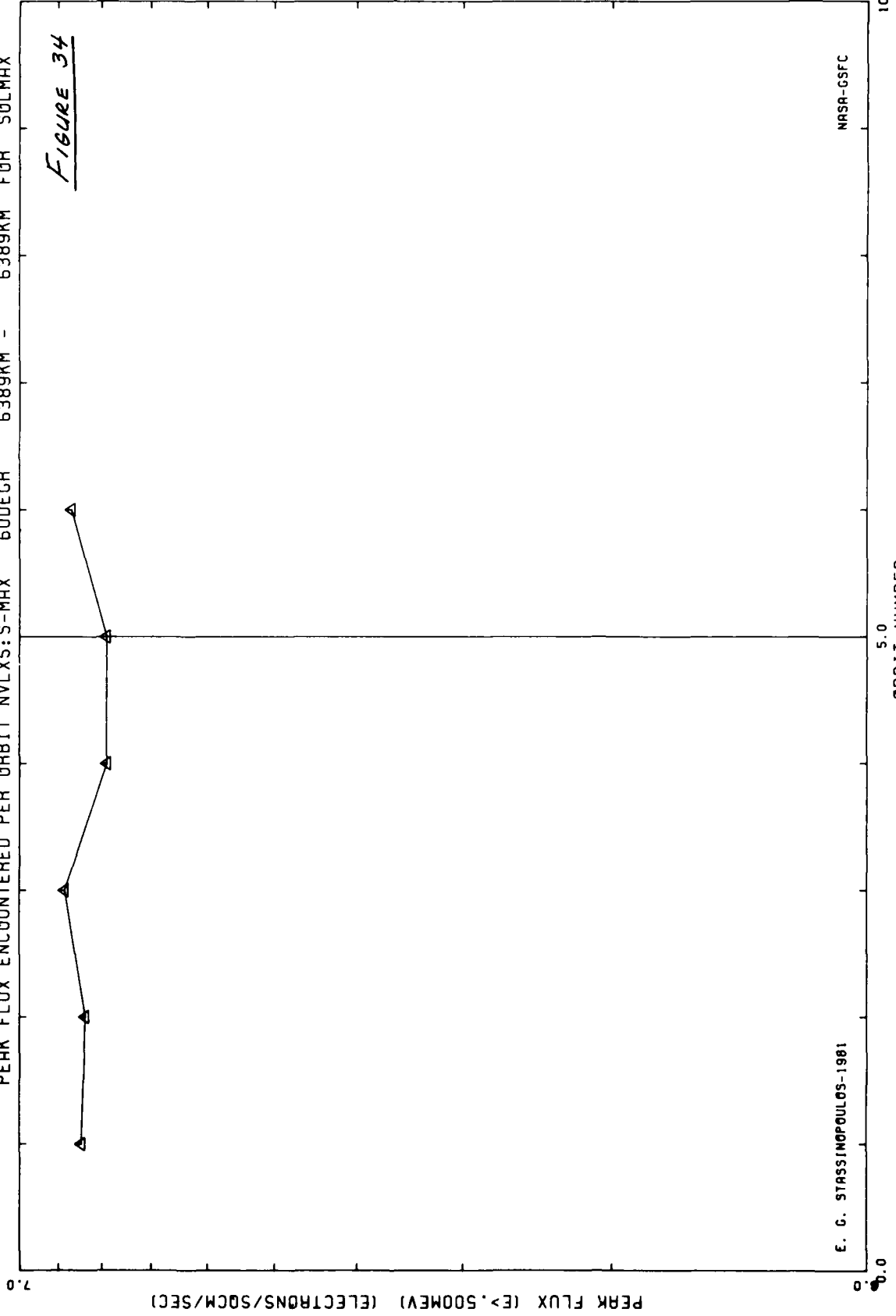
ORBIT NUMBER

7.0

PEAK FLUX (>S,0MEV) (PROTONS/SQCM/SEC)

PEAK FLUX ENCOUNTERED PER ORBIT NVLX5:S-MAX 60DEGR 6389KM - 6389KM FOR SOLMAX

FIGURE 34

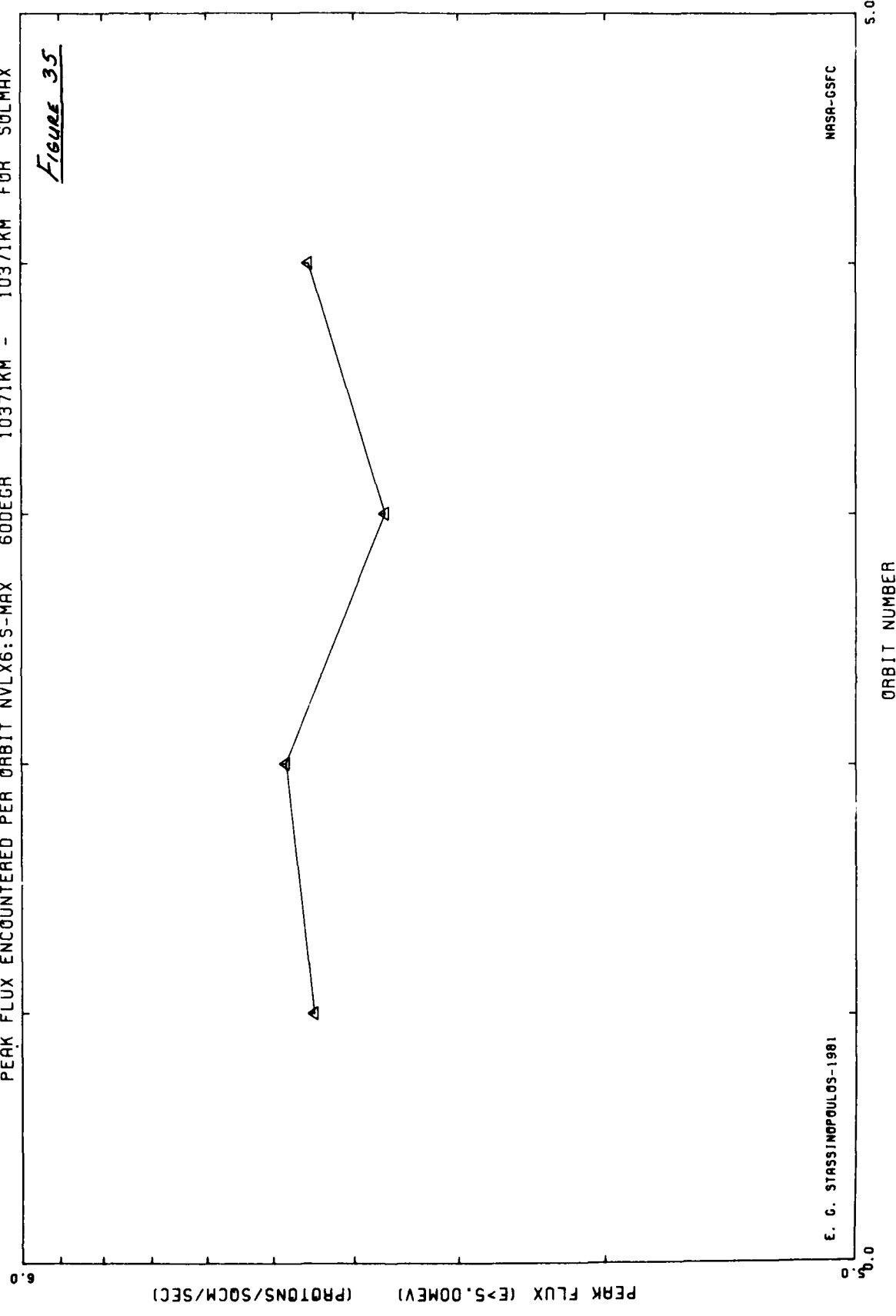


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PEAK FLUX ENCOUNTERED PER ORBIT NVLX6:S-MAX 60DEGR 10371KM - 10371KM FOR SOLMAX

Figure 35



E. C. STASSINOPOULOS-1981

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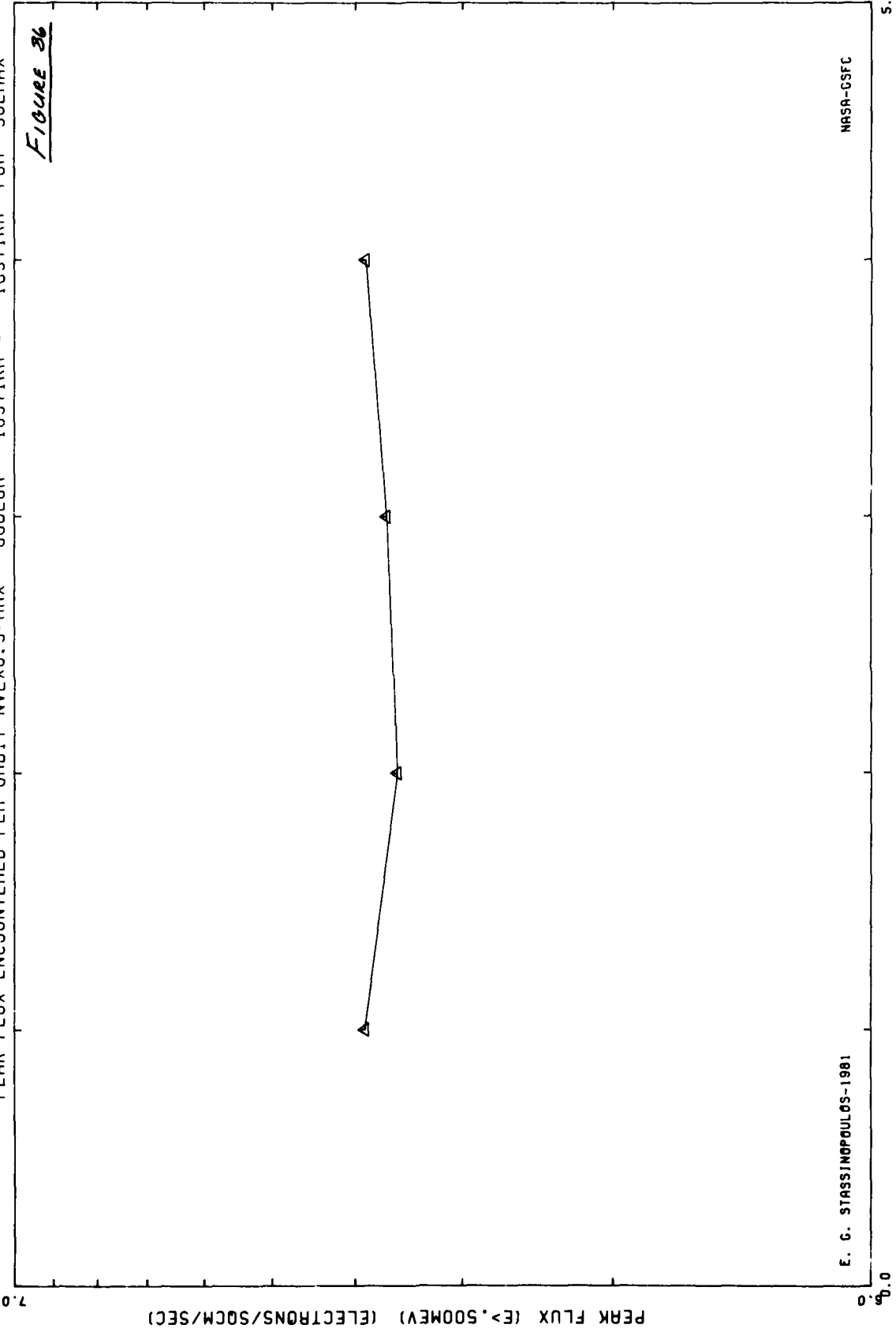
ORBIT NUMBER

0.0

5.0

PEAK FLUX ENCOUNTERED PER ORBIT NVLX6:S-MAX 60DEGR 10371KM - 10371KM FOR SOLMAX

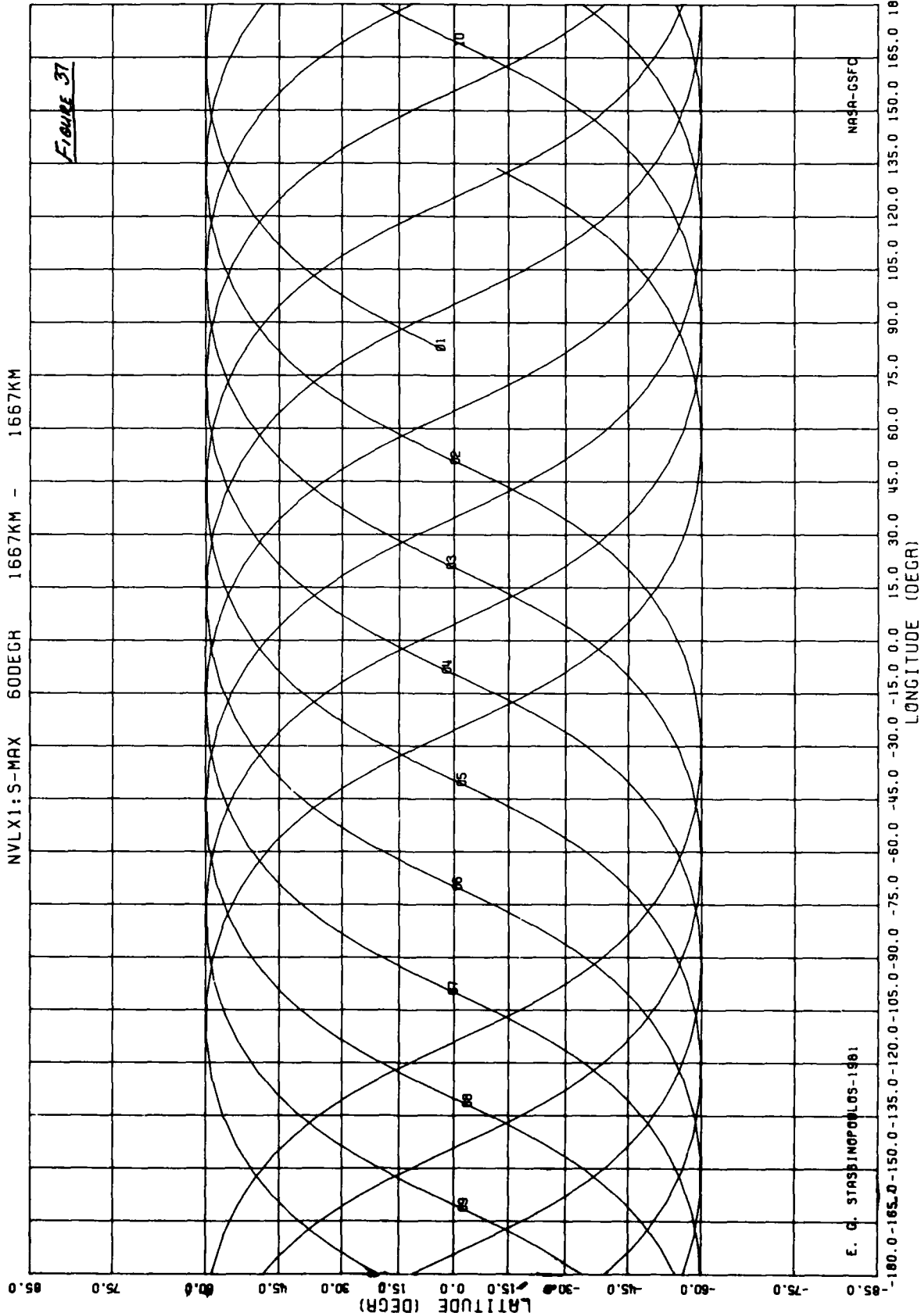
FIGURE 36



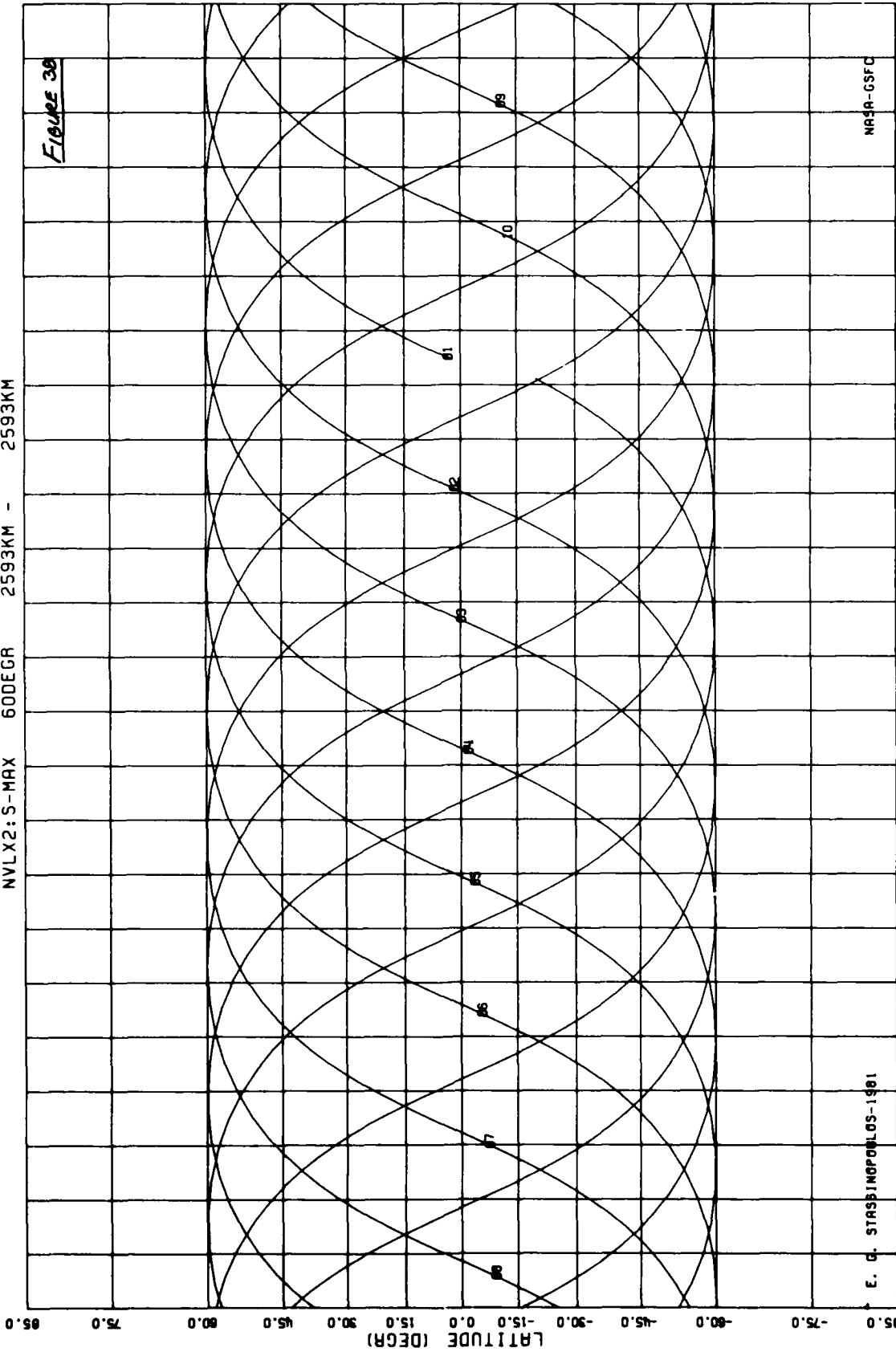
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NASA-CRFC

NVLX1:S-MAX 60DEGH 1667KM - 1667KM



NVLX2:S-MAX 60DEGR 2593KM - 2593KM

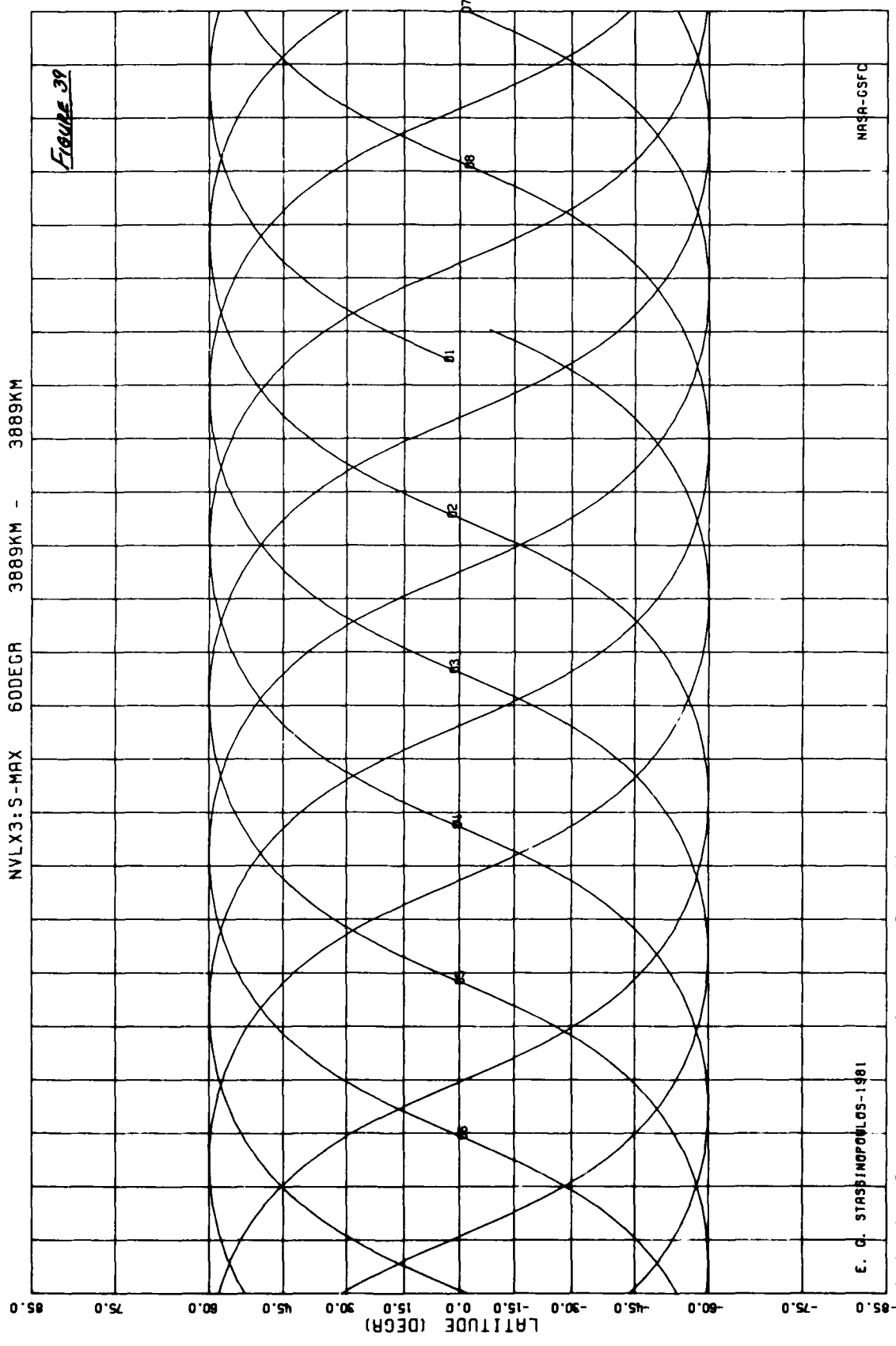


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85.0  
-85.0  
-75.0  
-60.0  
-45.0  
-30.0  
-15.0  
0.0  
15.0  
30.0  
45.0  
60.0  
75.0  
90.0  
105.0  
120.0  
135.0  
150.0  
165.0  
180.0  
LONGITUDE (DEGR)





NVLX3: S-MAX 60DEGR 3889KM - 3889KM

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LATITUDE (DEGR)

LONGITUDE (DEGR)

-85.0 -75.0 -60.0 -45.0 -30.0 -15.0 0.0 15.0 30.0 45.0 60.0 75.0 90.0 105.0 120.0 135.0 150.0 165.0 180.0

80

81

82

83

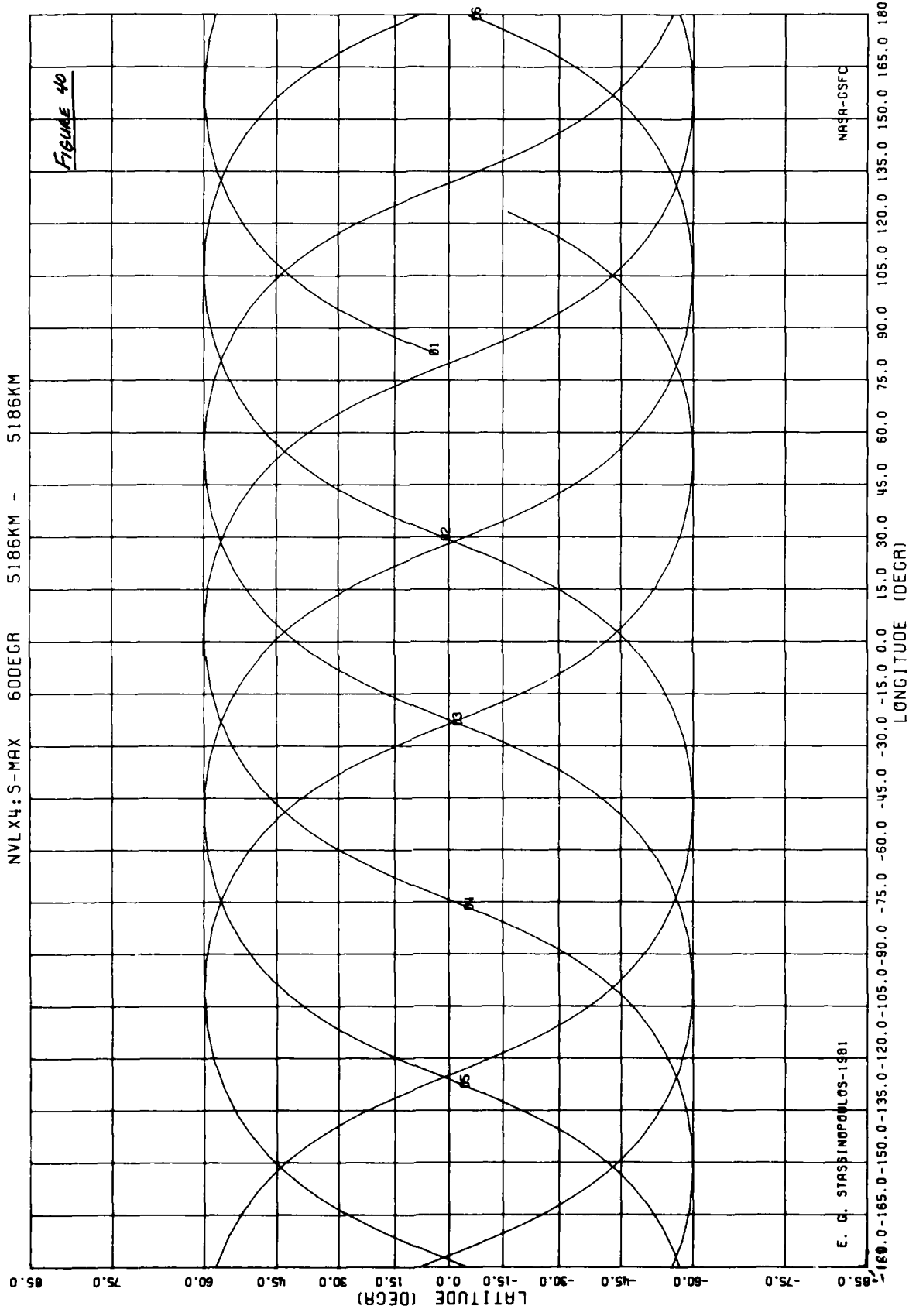
84

85

86

87

88



NVLX5:S-MAX 60DEGR 6389KM - 6389KM

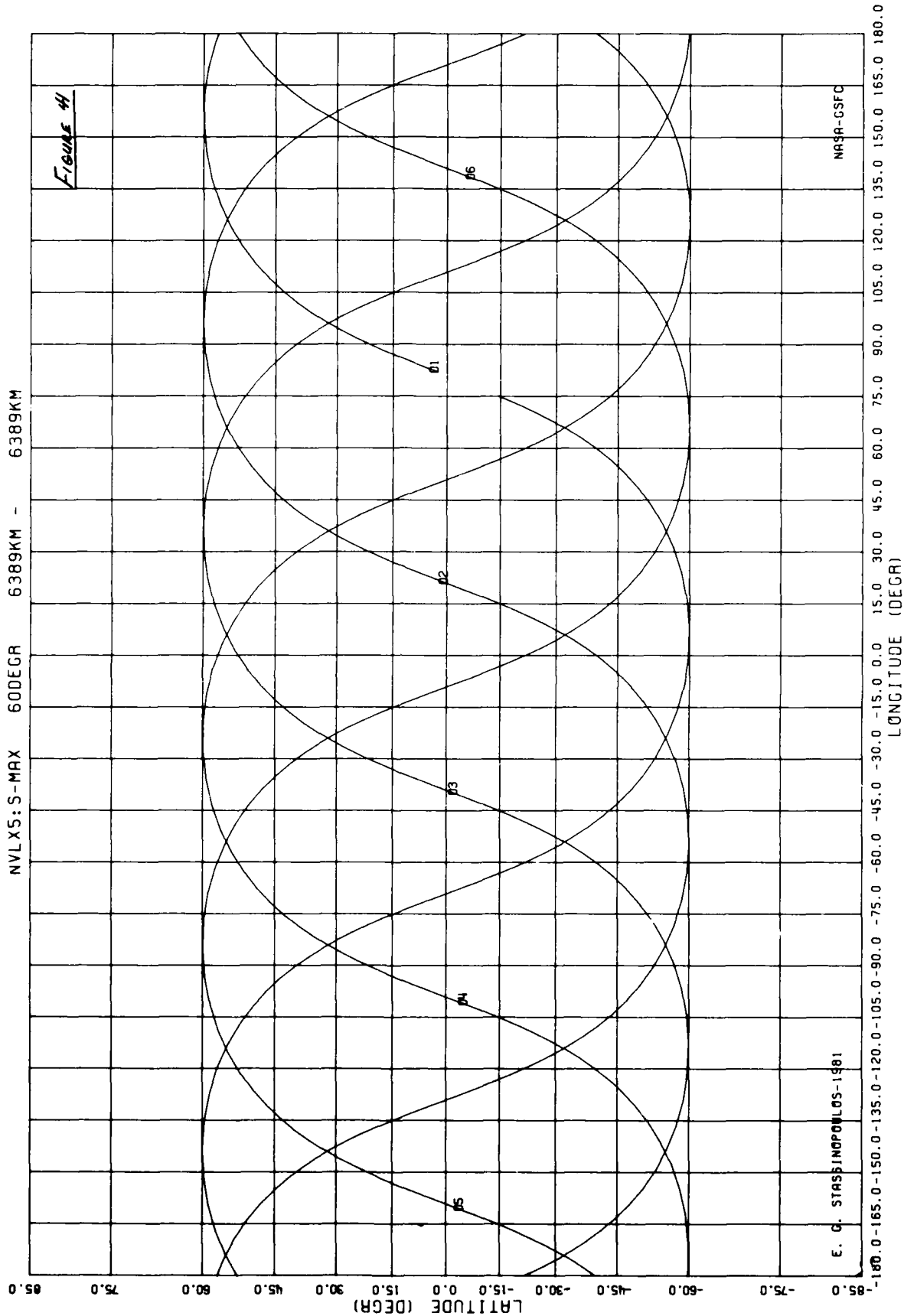
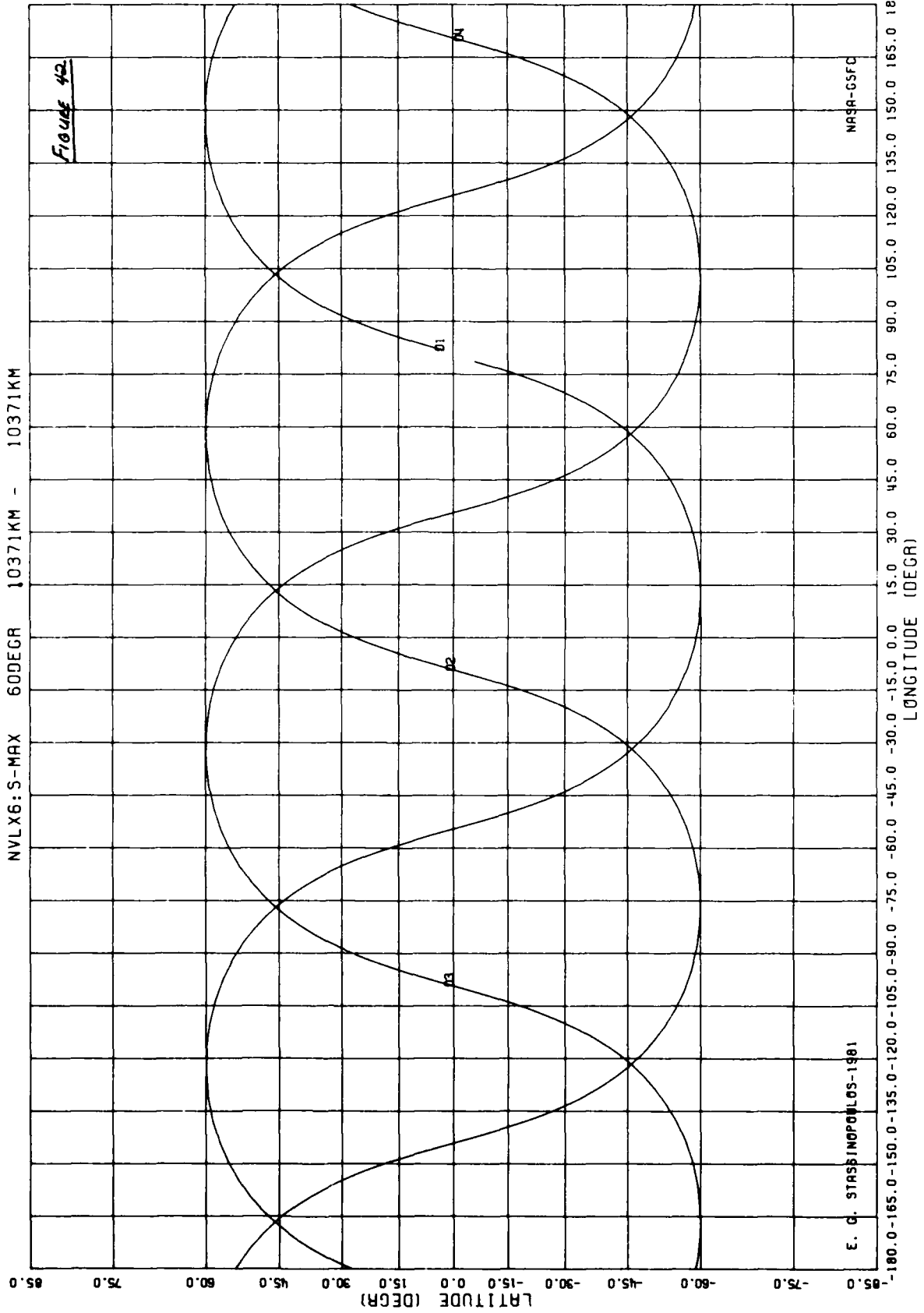


FIGURE 4

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NARS-CSTC

NVLX6: S-MAX 60DEGR 10371KM - 10371KM

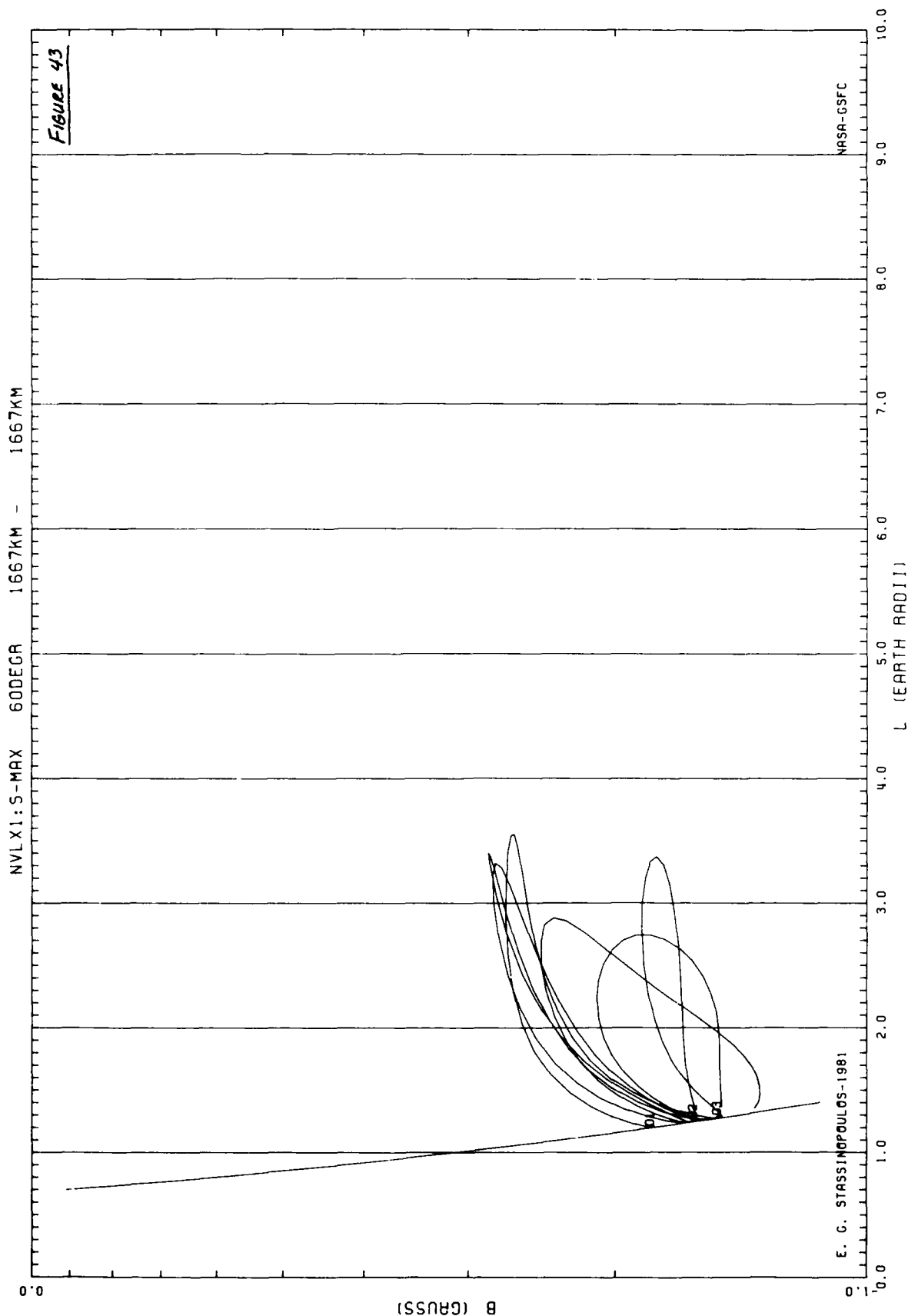


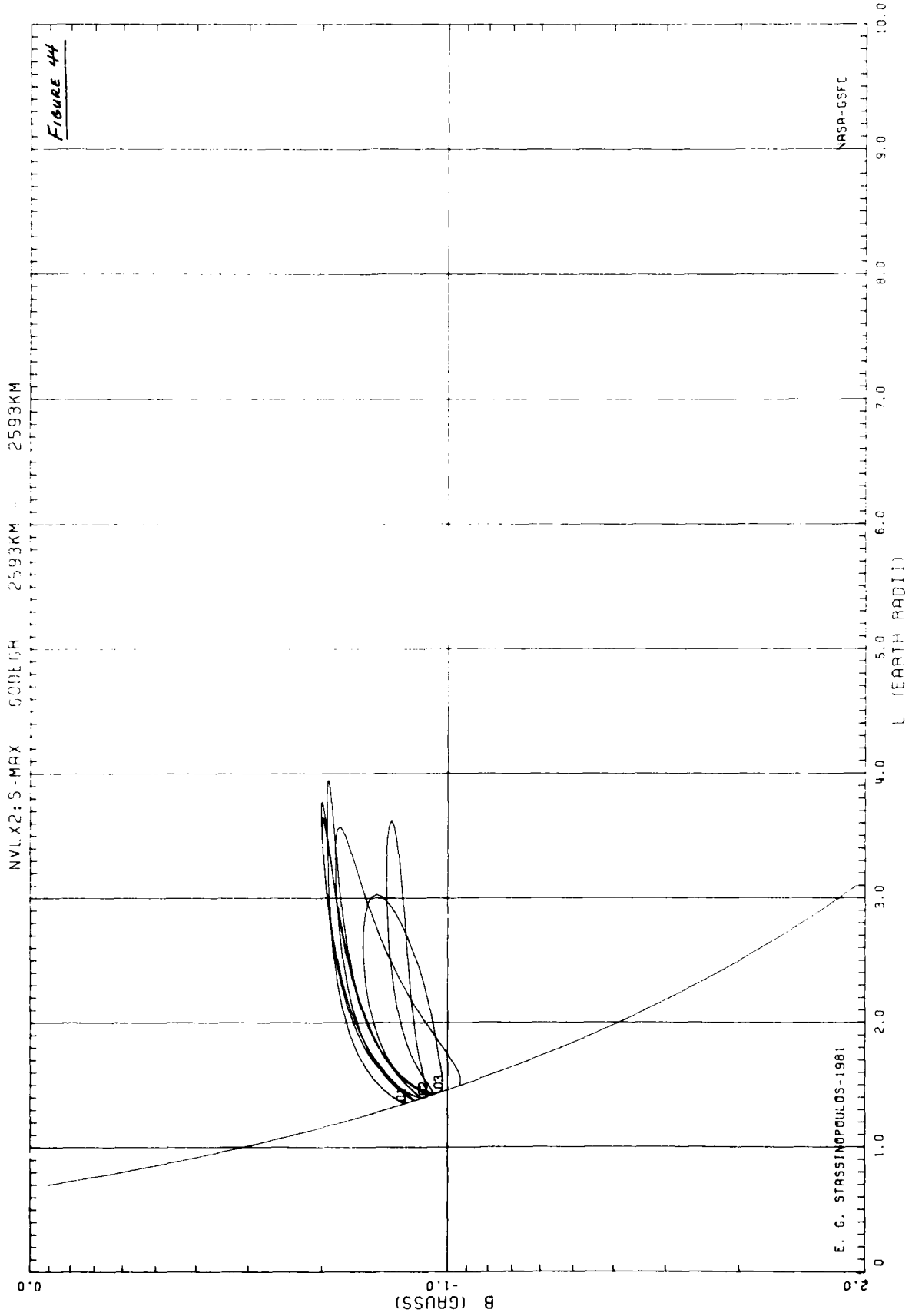
E. G. STASSINOPoulos-1981

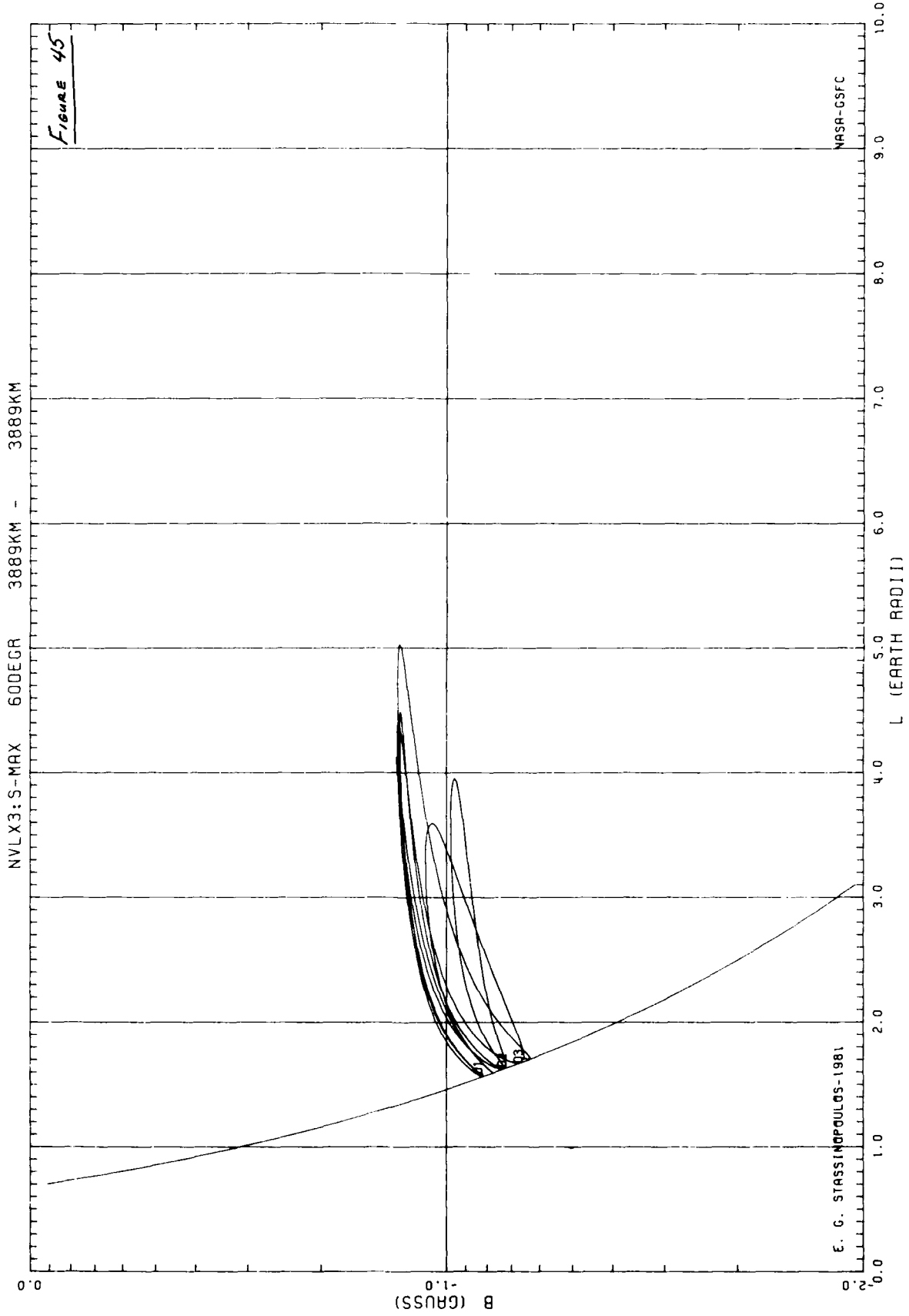
NASA-CSFC

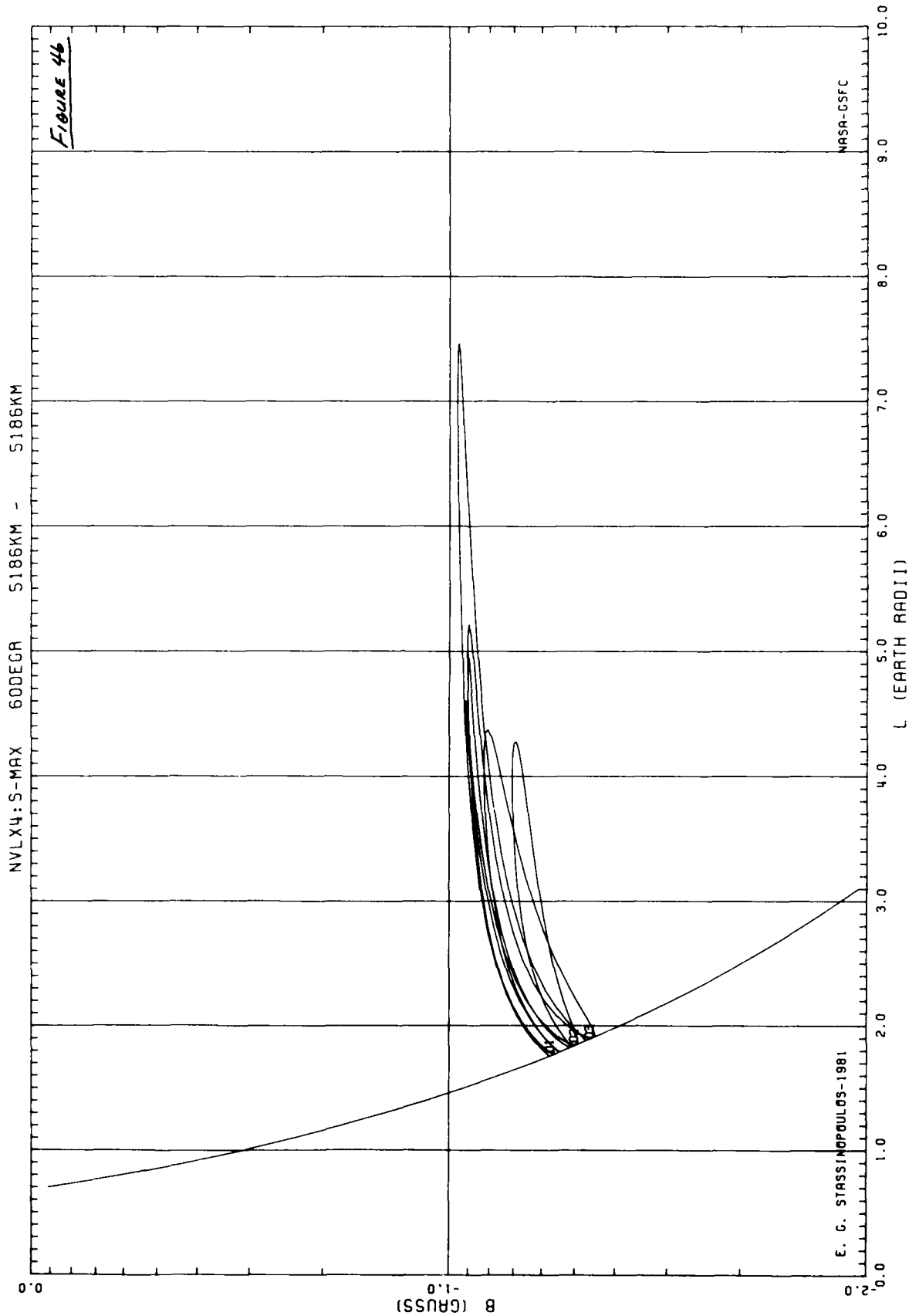
85.0  
-85.0  
-75.0  
-60.0  
-45.0  
-30.0  
-15.0  
0.0  
15.0  
30.0  
45.0  
60.0  
75.0  
90.0  
105.0  
120.0  
135.0  
150.0  
165.0  
180.0

LONGITUDE (DEGR)

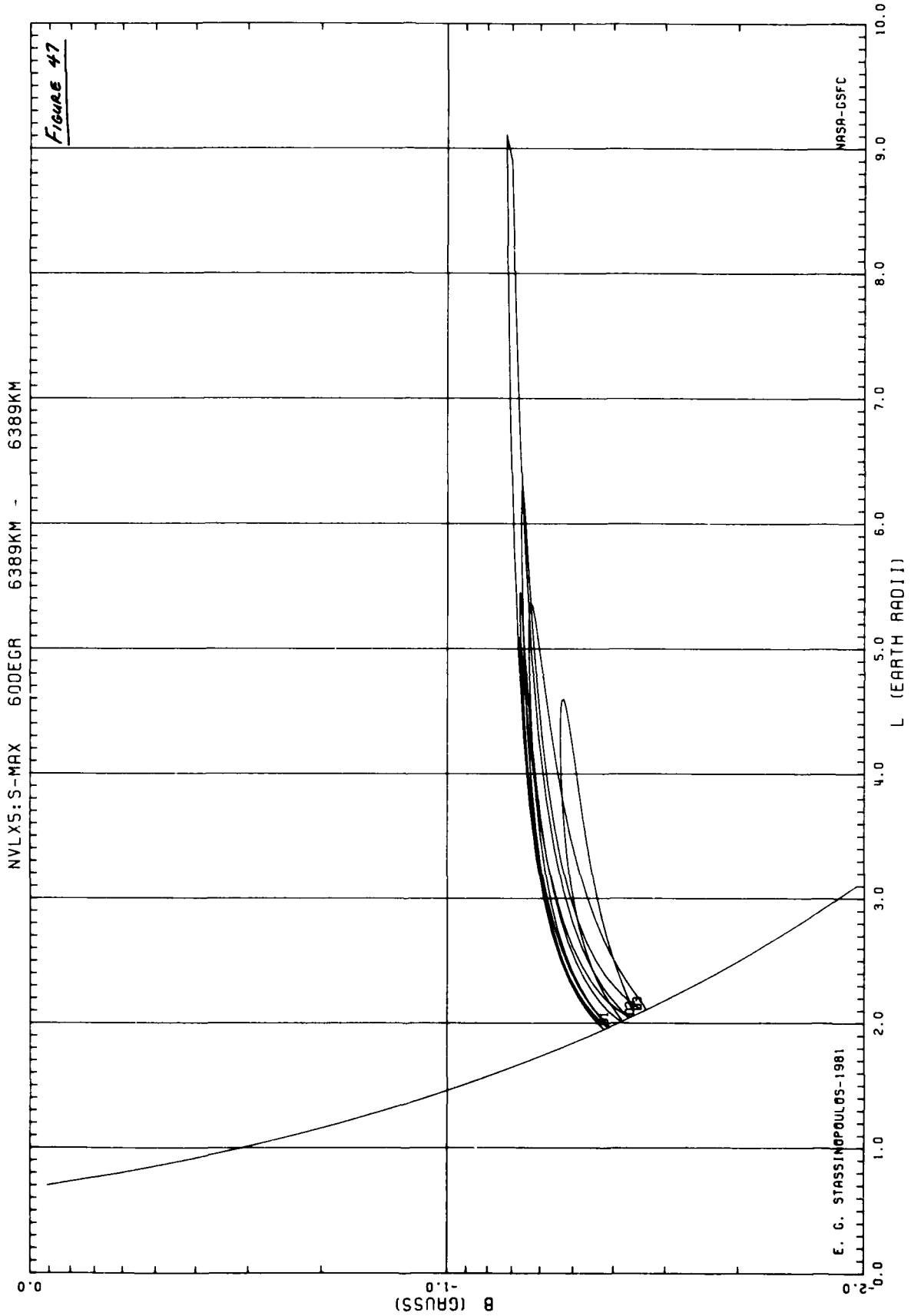


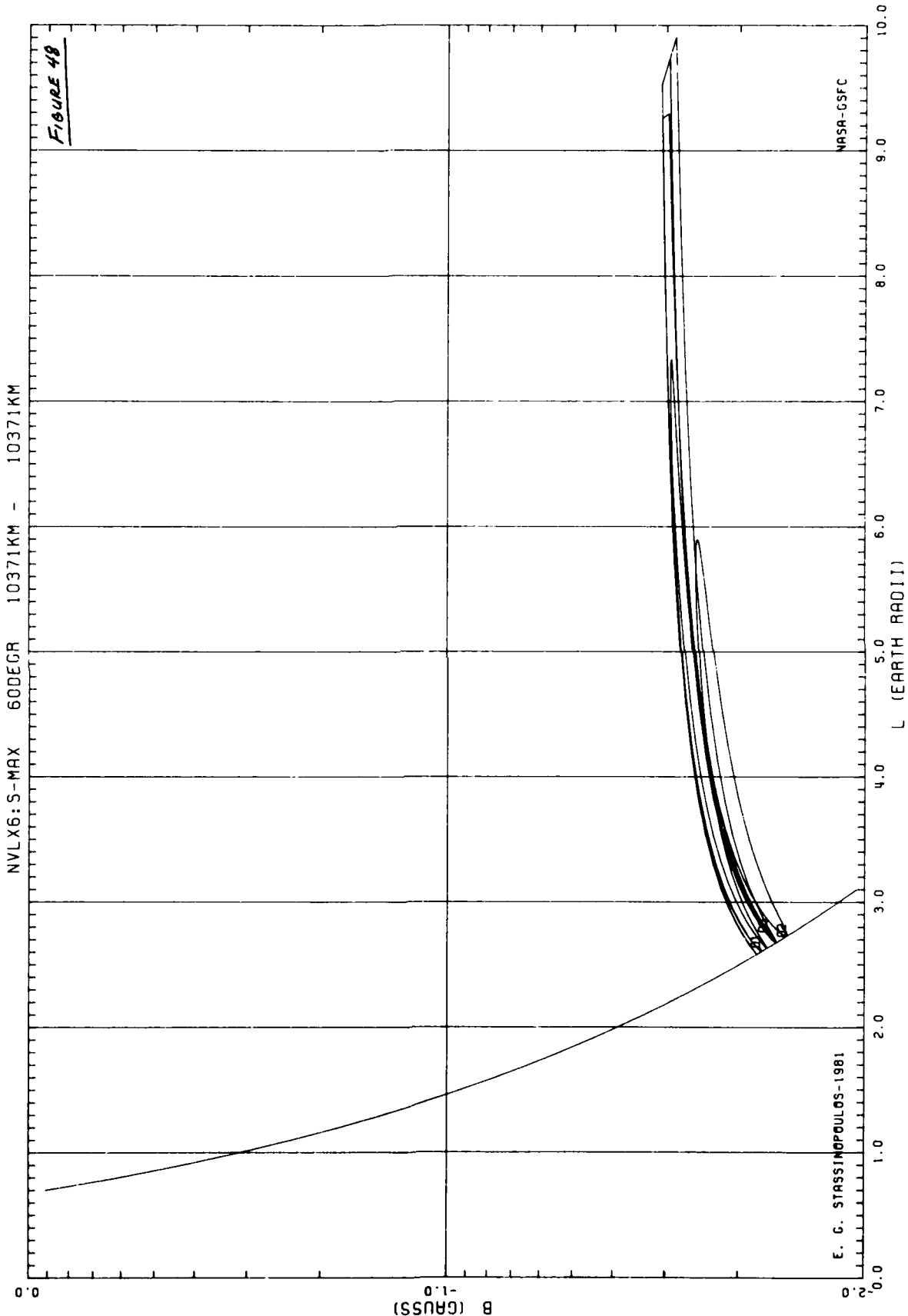






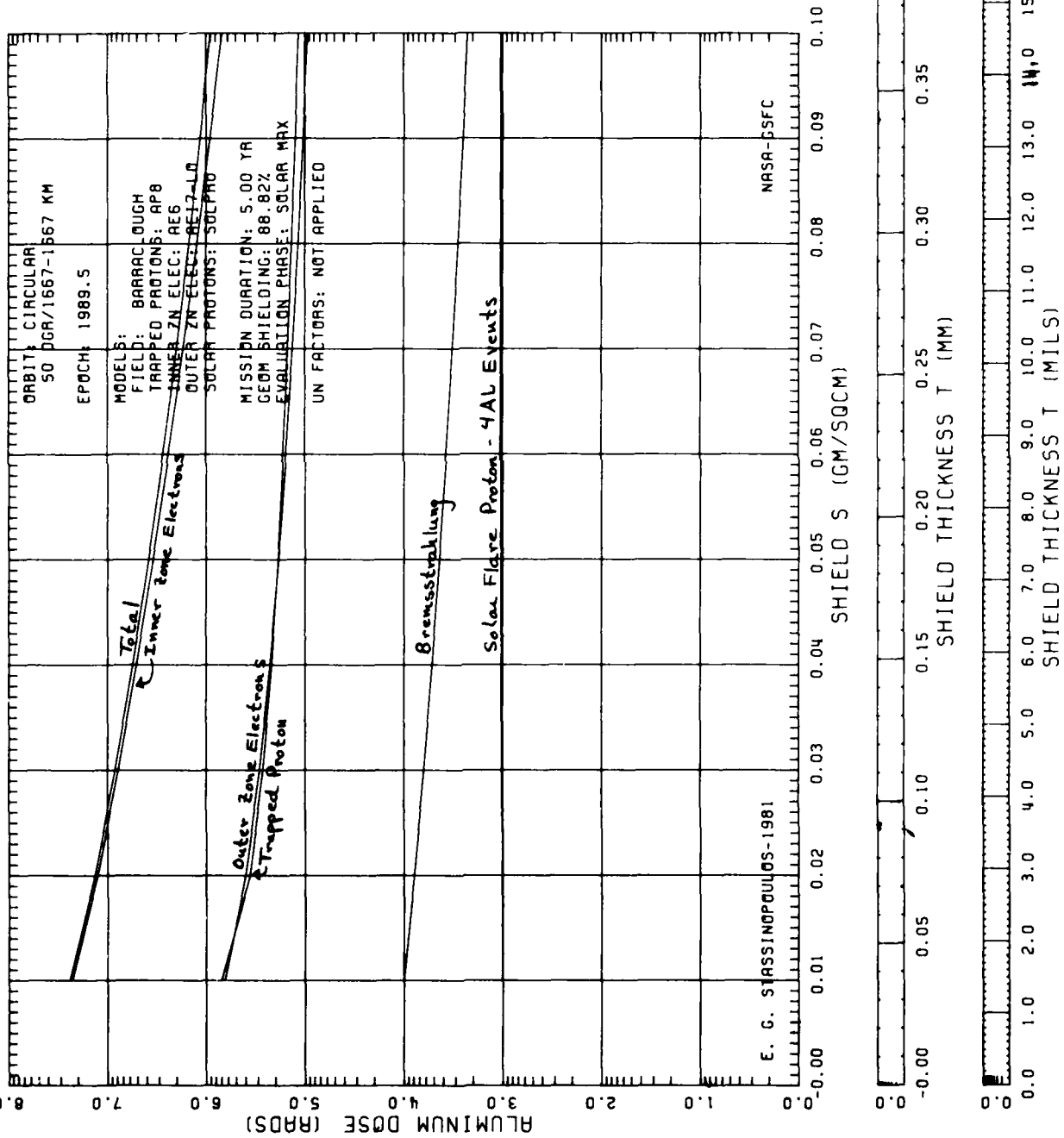






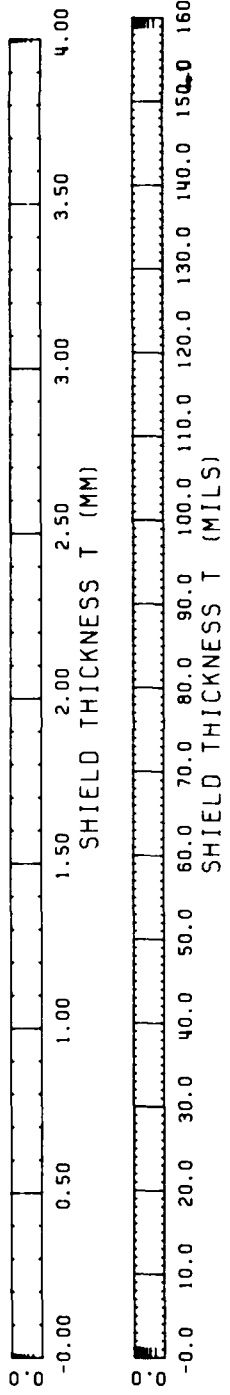
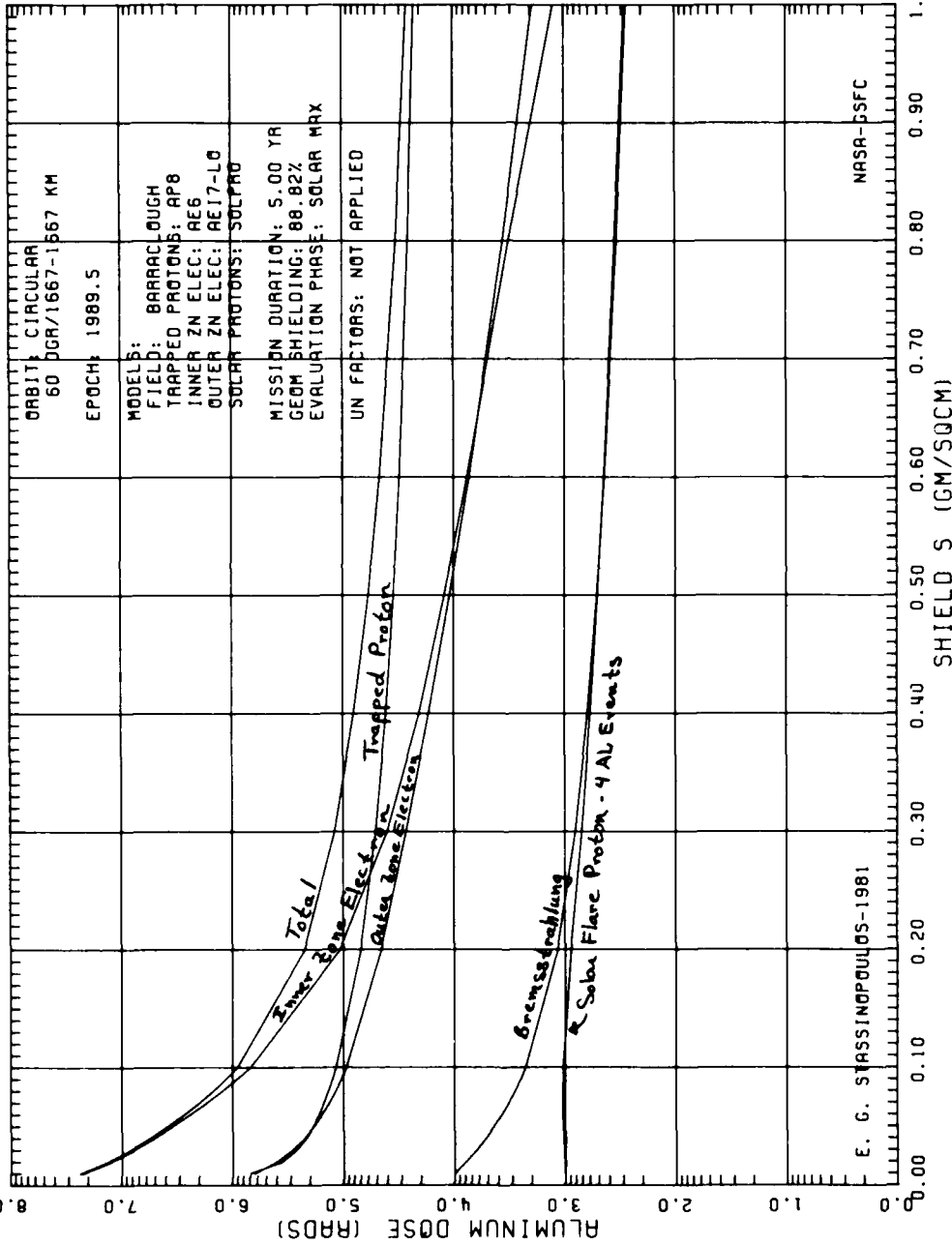
**FIGURE 49**

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLXI:S-MAX



**Figure 50**

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX1: S-MAX



**FIGURE 51**

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX1: S-MAX

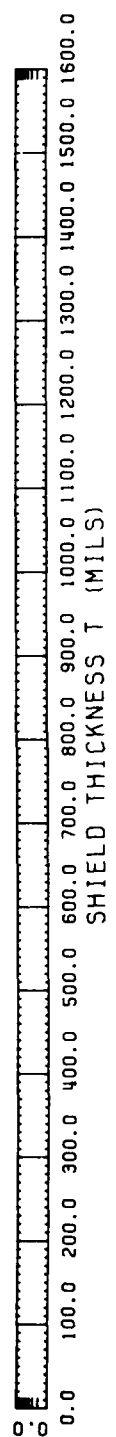
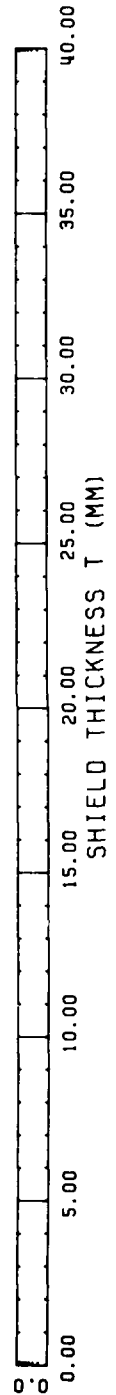
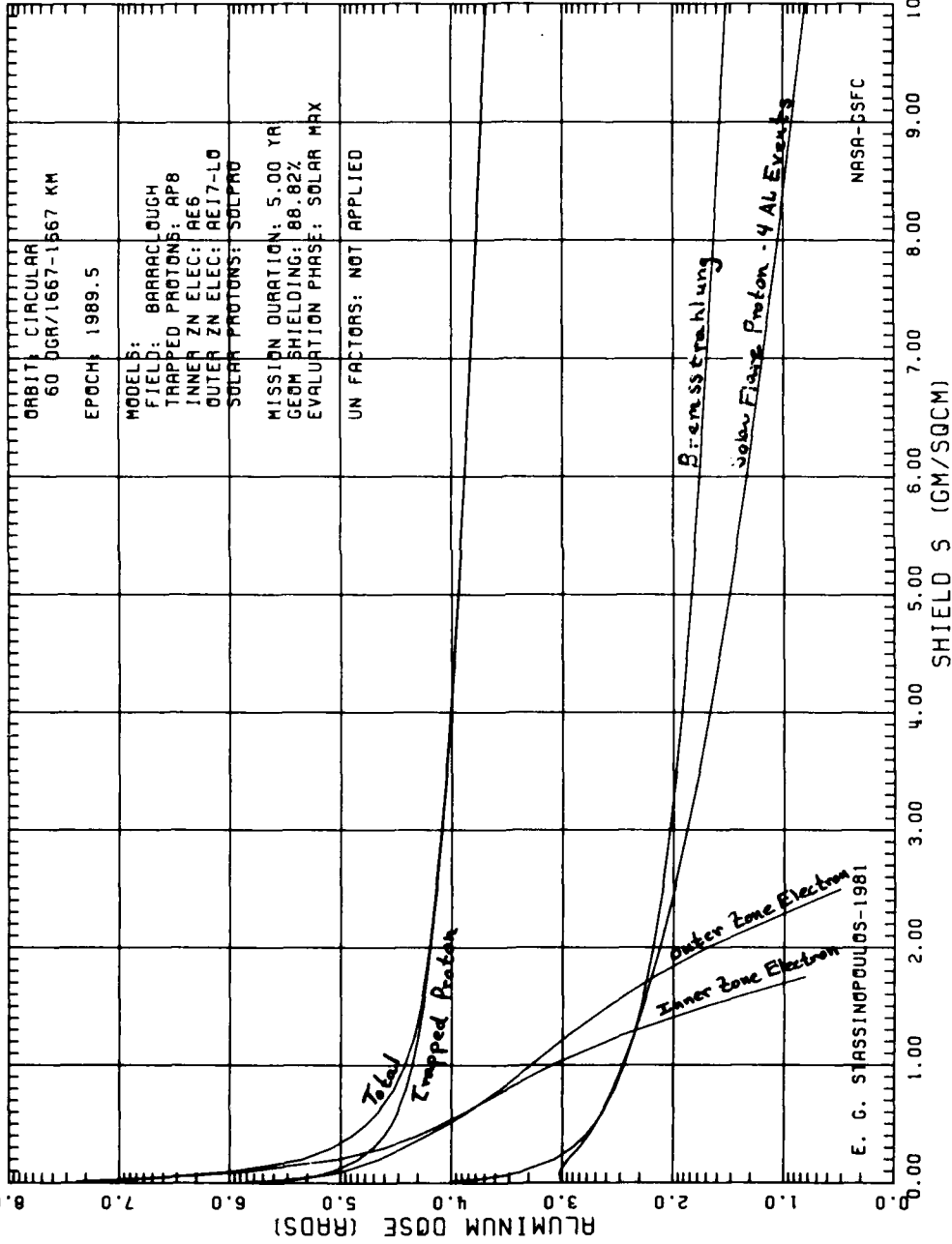


FIGURE 52

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX1:S-MAX

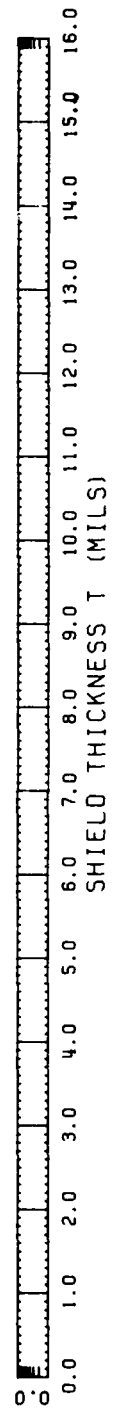
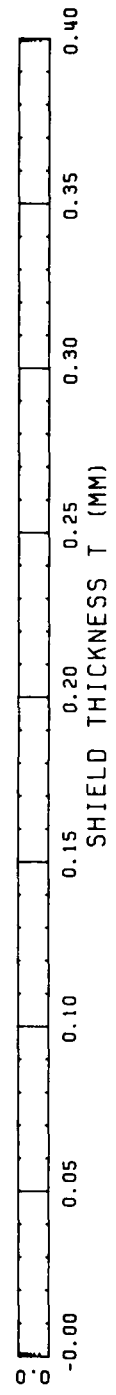
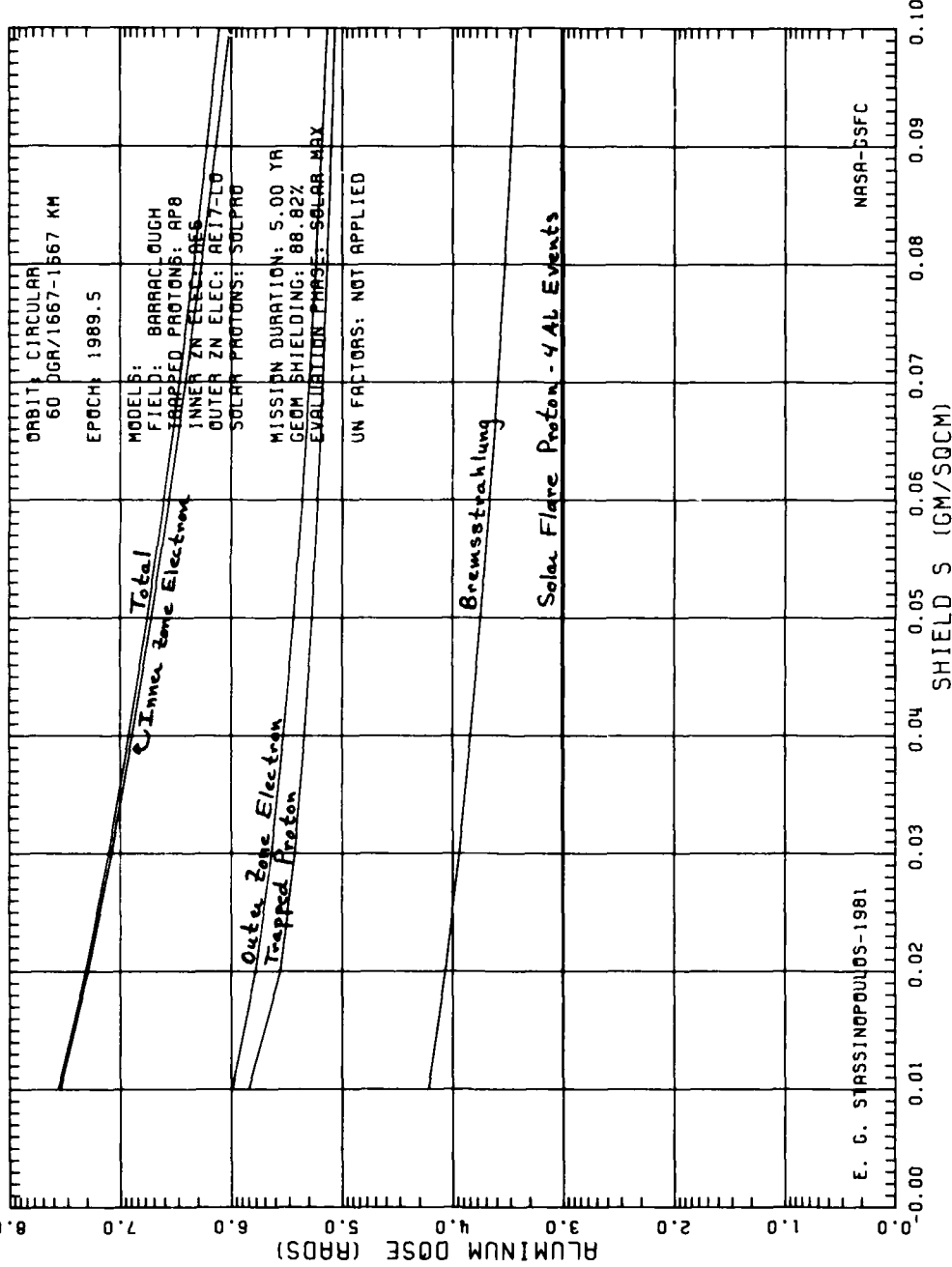


Figure 53

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX1:S-MAX

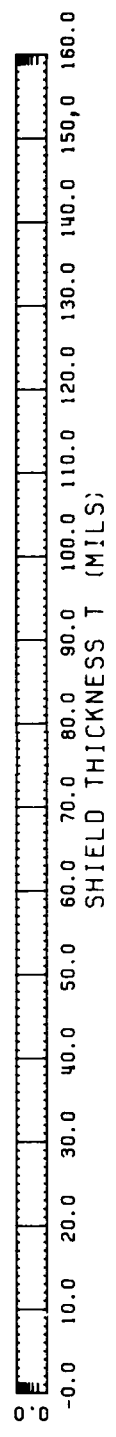
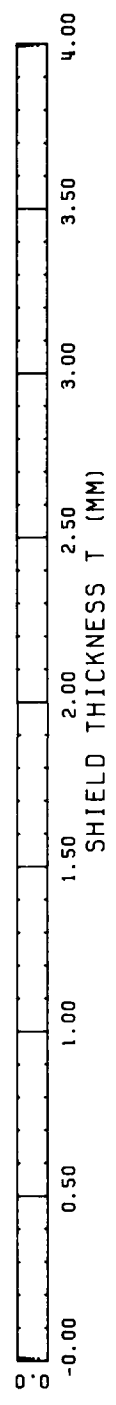
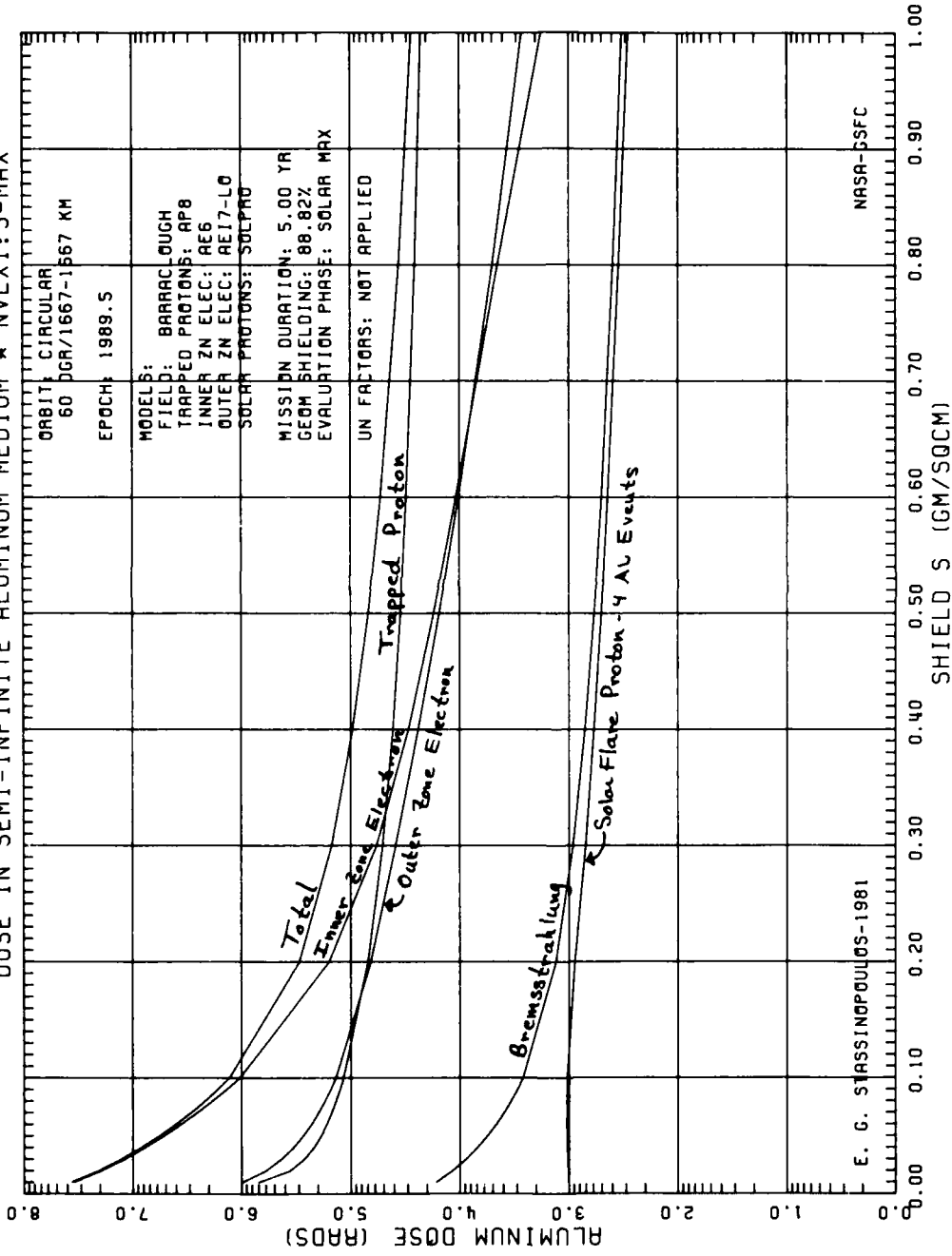
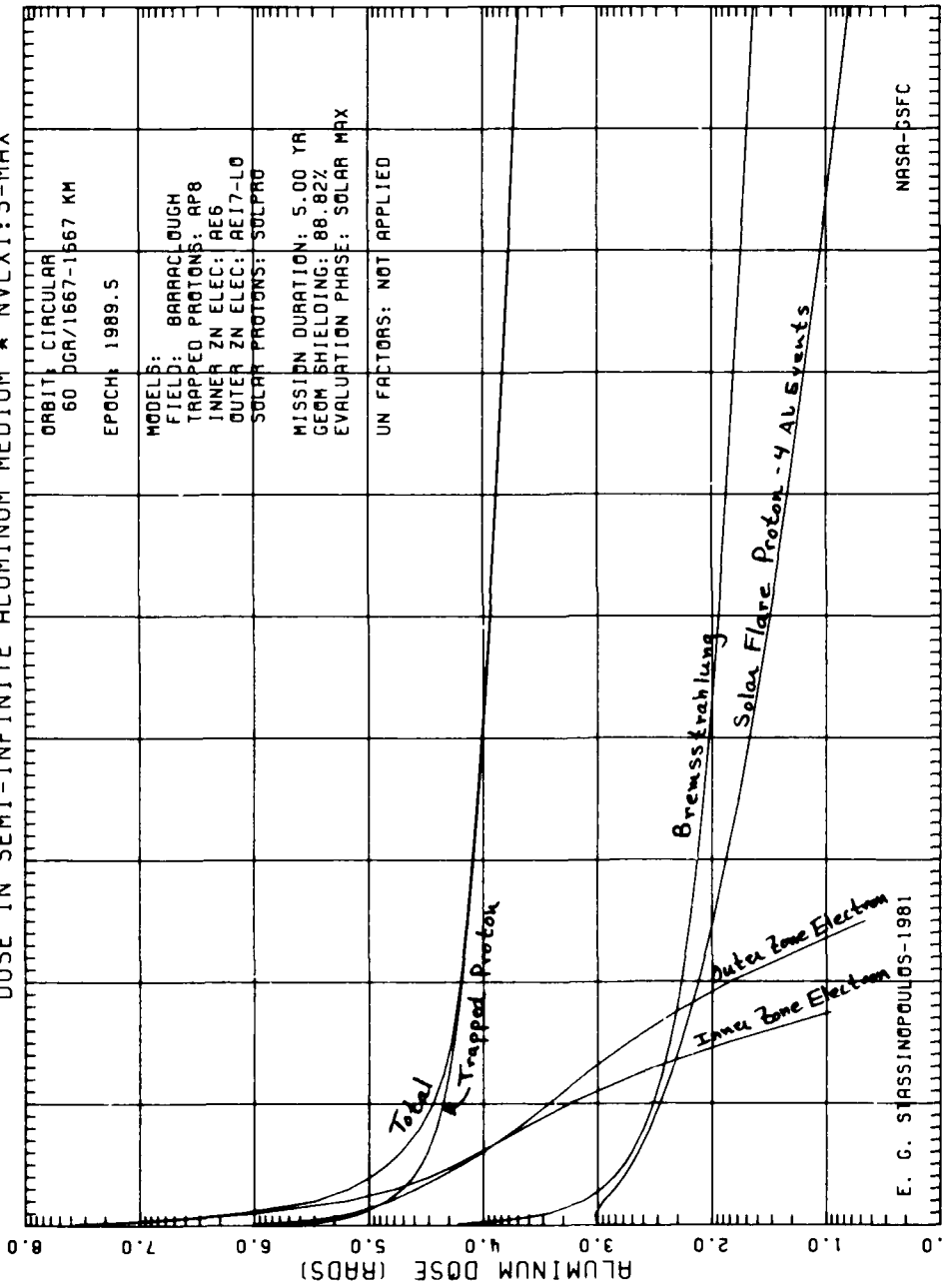


Figure 54

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX1:S-MAX



ORBIT: CIRCULAR  
60 DGR/1667-1667 KM  
EPOCH: 1989.5

MODELS:  
FIELD: BARRACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AE17-LO  
SOLAR PROTONS: SOLPRO

MISSION DURATION: 5.00 YA  
GEOM SHIELDING: 88.82%  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

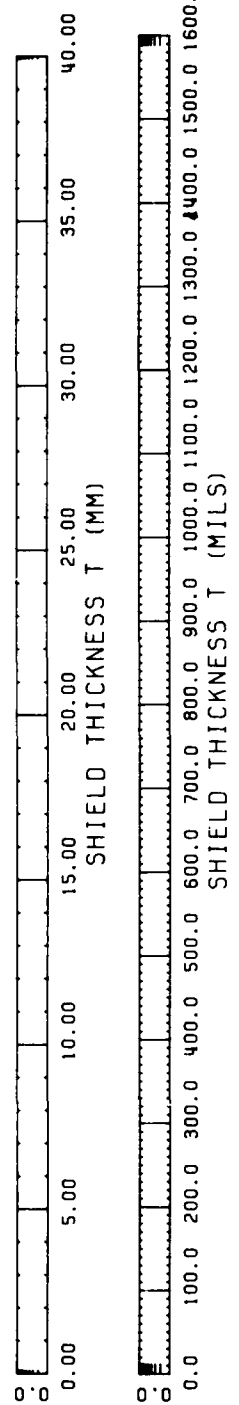




FIGURE 55

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX1:S-MAX

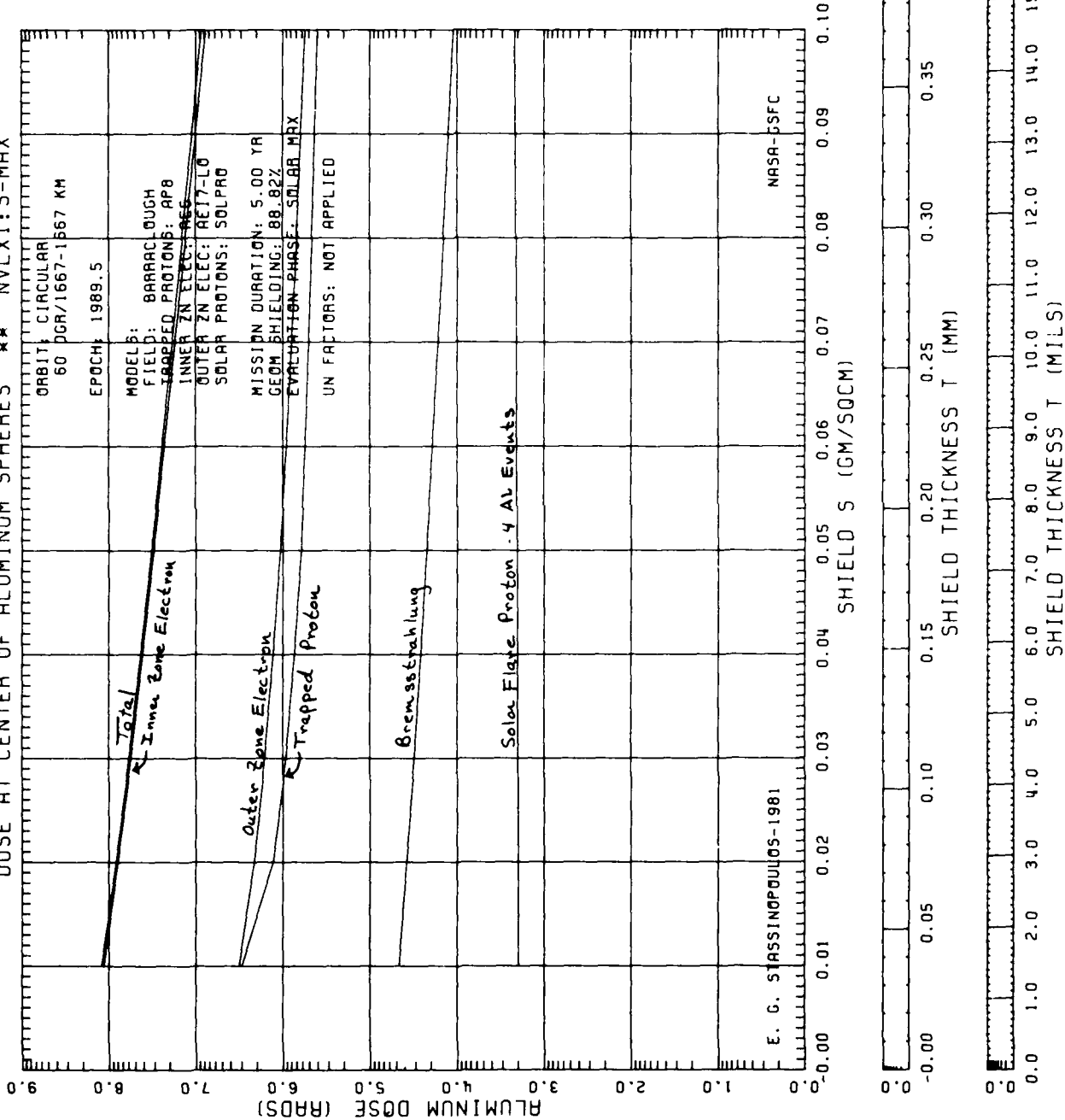


Figure 56

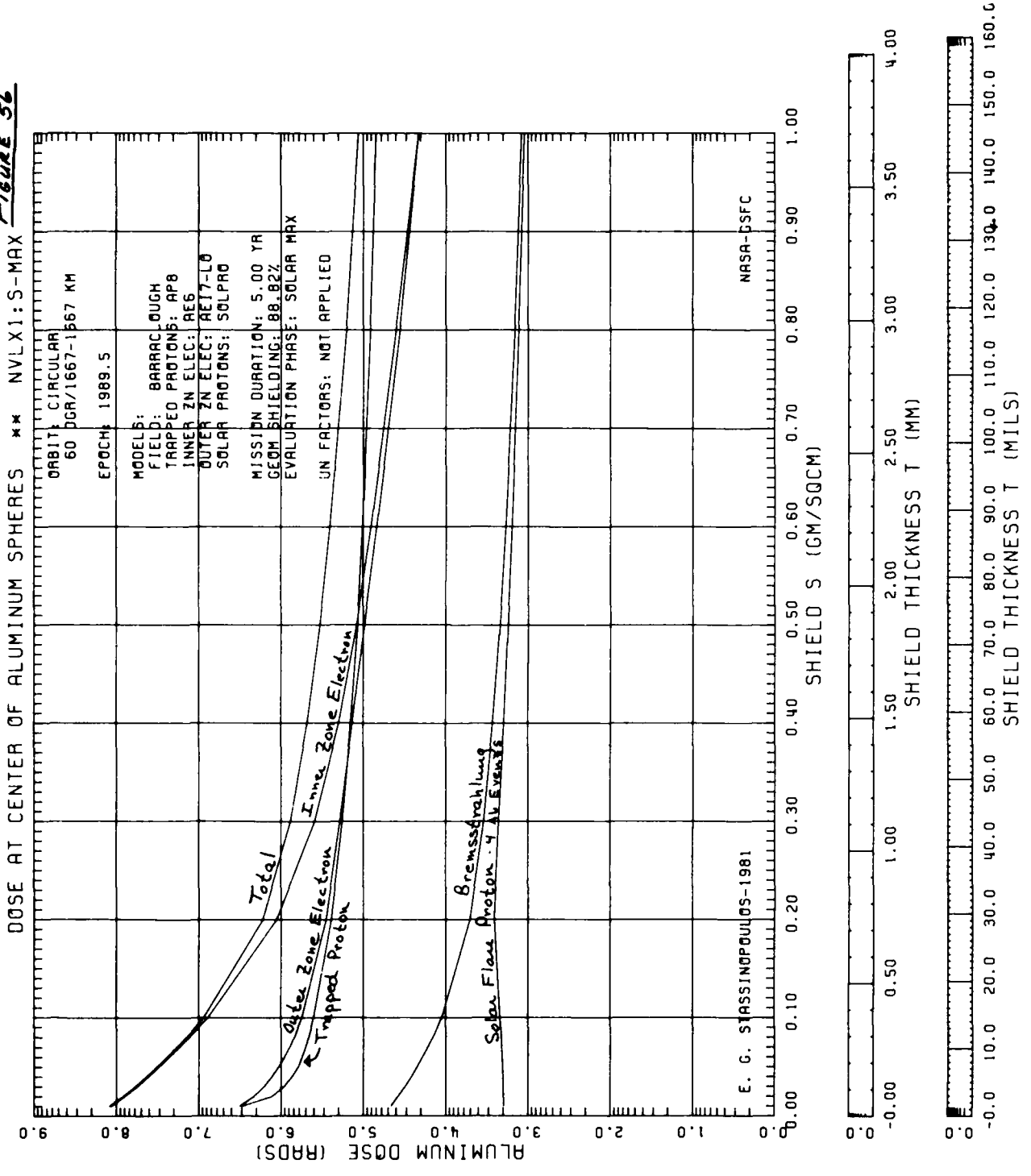
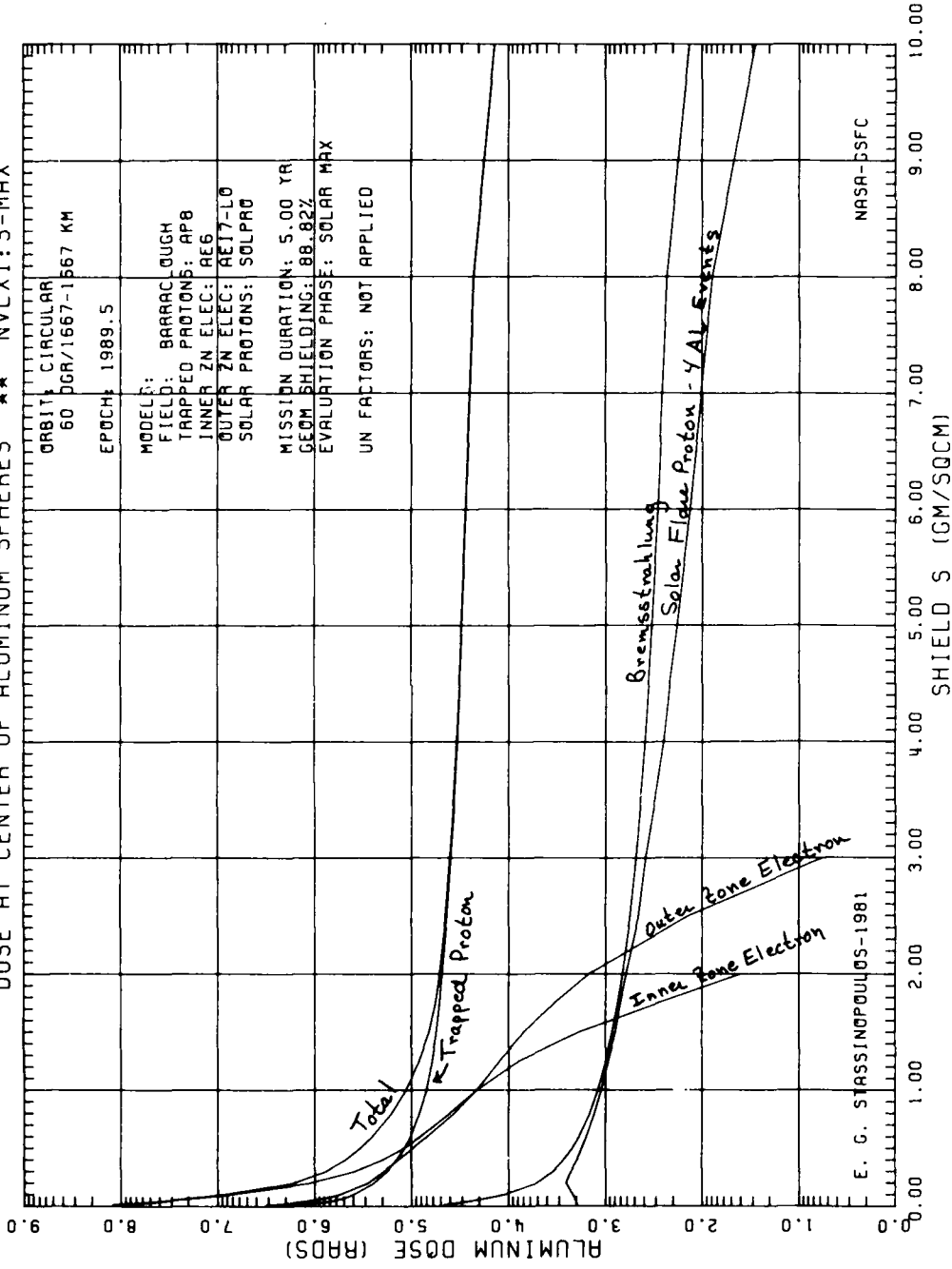
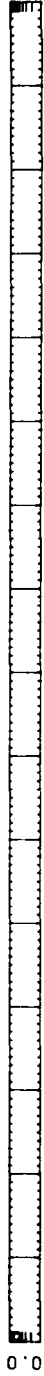
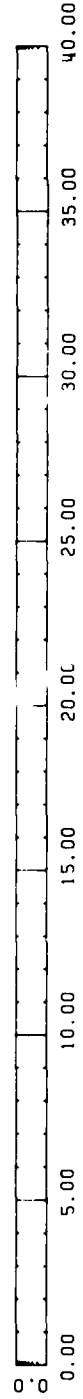


FIGURE 57

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX1:S-MAX



E. G. STASSINOPoulos-1981



ORBIT: CIRCULAR  
60 DGR/1667-1867 KM

EPOCH: 1989.5

MODEL: BARRACLOUGH  
FIELD: TRAPPED PROTONS: APB  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AET7-L0  
SOLAR PROTONS: SOLPA0

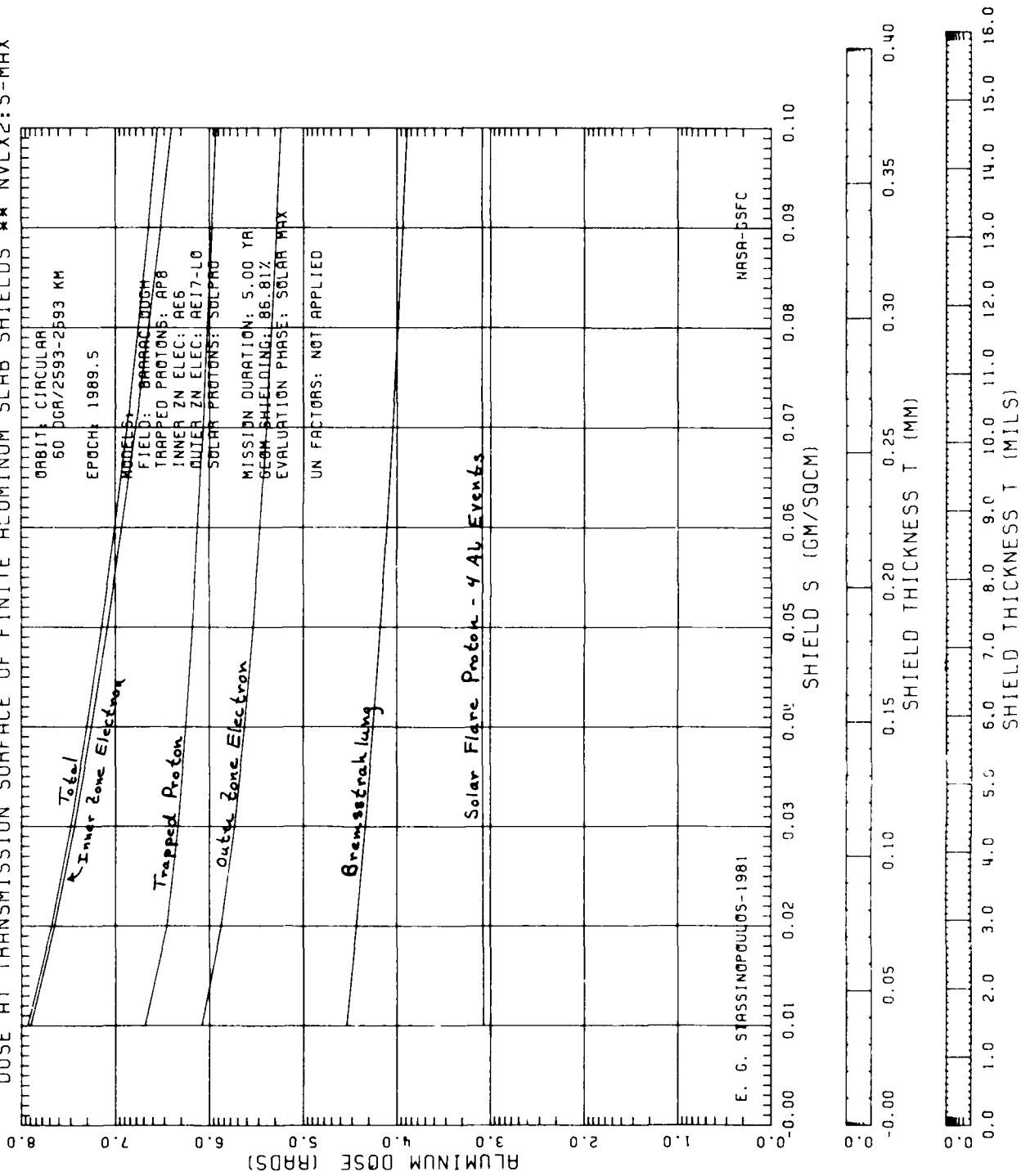
MISSION DURATION: 5.00 YR  
GEM SHIELDING: 88.82%

EVALUATION PHASE: SOLAR MAX

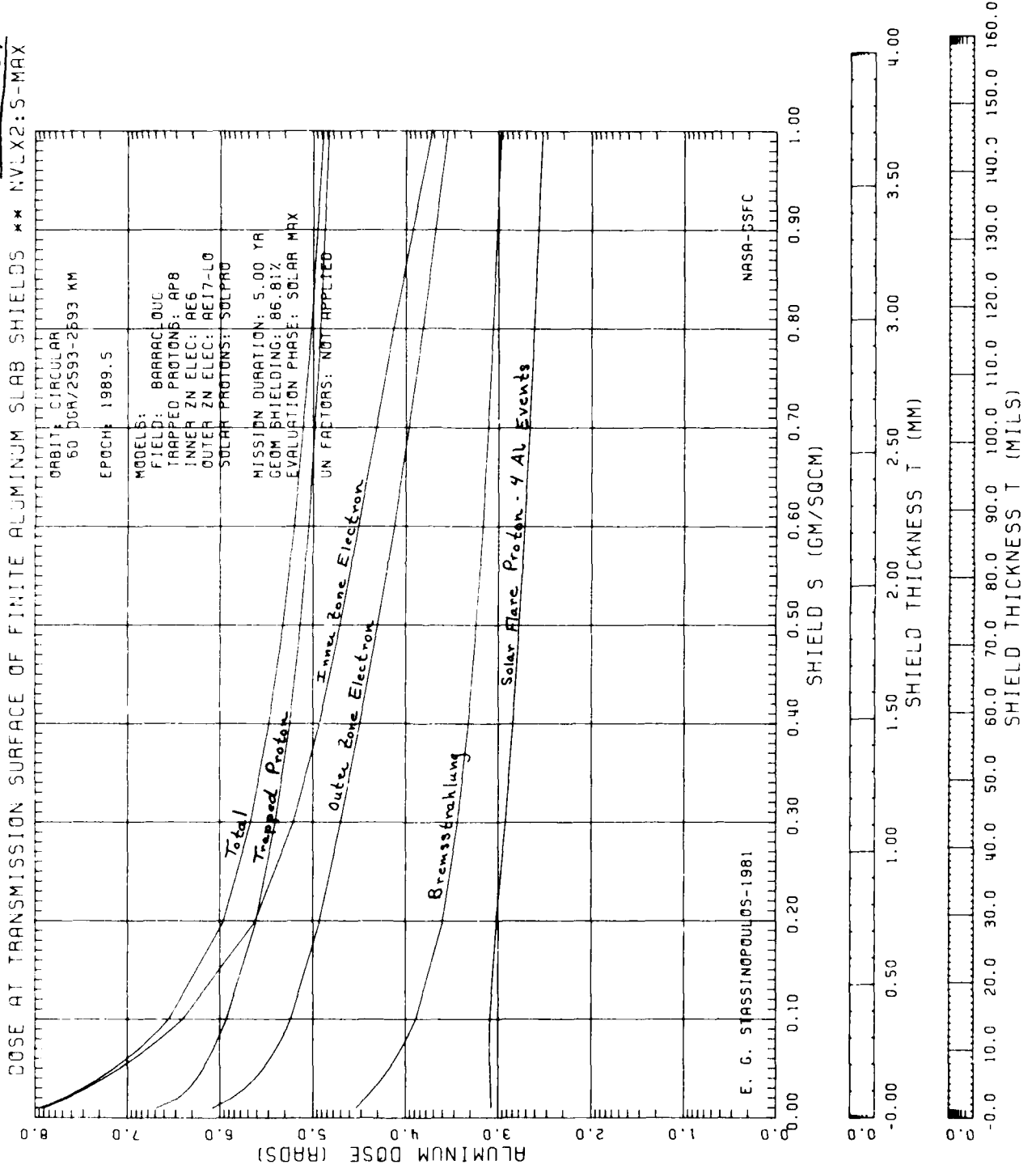
UN FACTORS: NOT APPLIED

0.0 100.0 200.0 300.0 400.0 500.0 600.0 700.0 800.0 900.0 1000.0 1100.0 1200.0 1300.0 1400.0 1500.0 1600.0

**FIGURE 50**  
**DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX2:S-MAX**

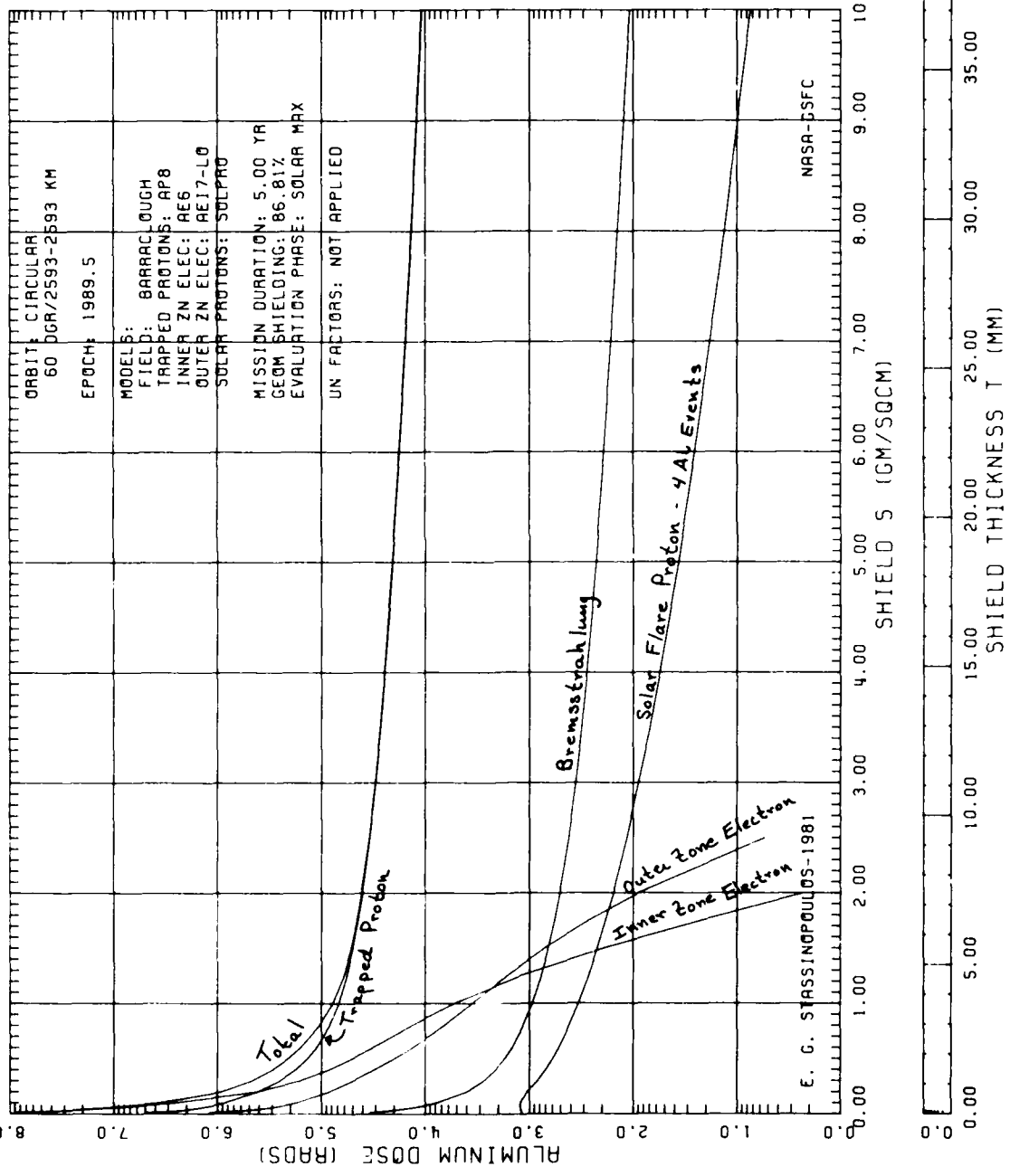


**FIGURE 59**  
**NVLX2:S-MAX**



**FIGURE 60**

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX2: S-MAX



ORBIT: CIRCULAR  
60 DGR/2593-2593 KM  
EPOCH: 1989.5

MODELS:  
FIELD: BARRACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
SOLAR PROTONS: SOLPRO

MISSION DURATION: 5.00 YR  
GEM SHIELDING: 86.81%  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

FIGURE 61

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX2: S-MAX

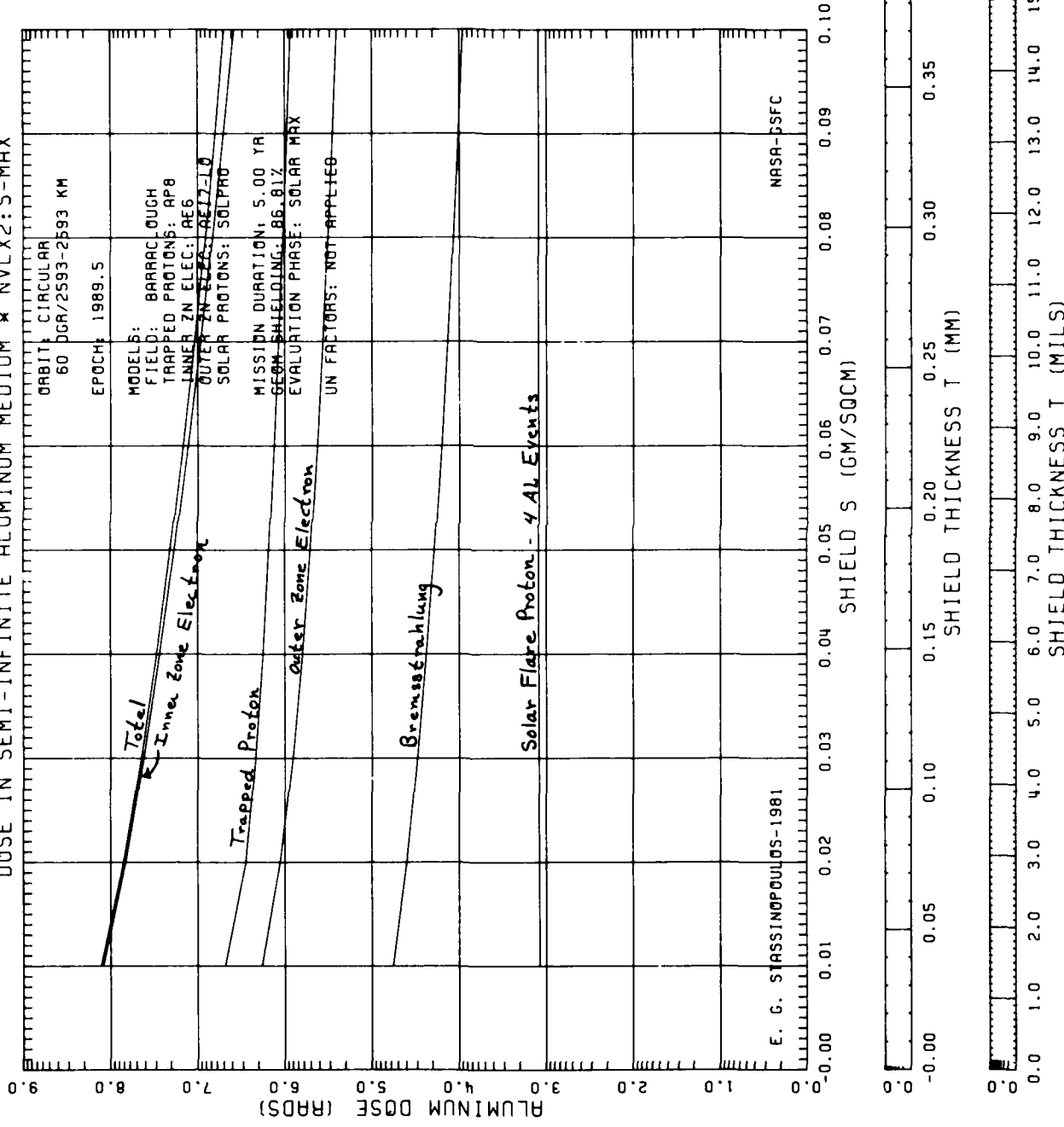


FIGURE 62

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX2:S-MAX

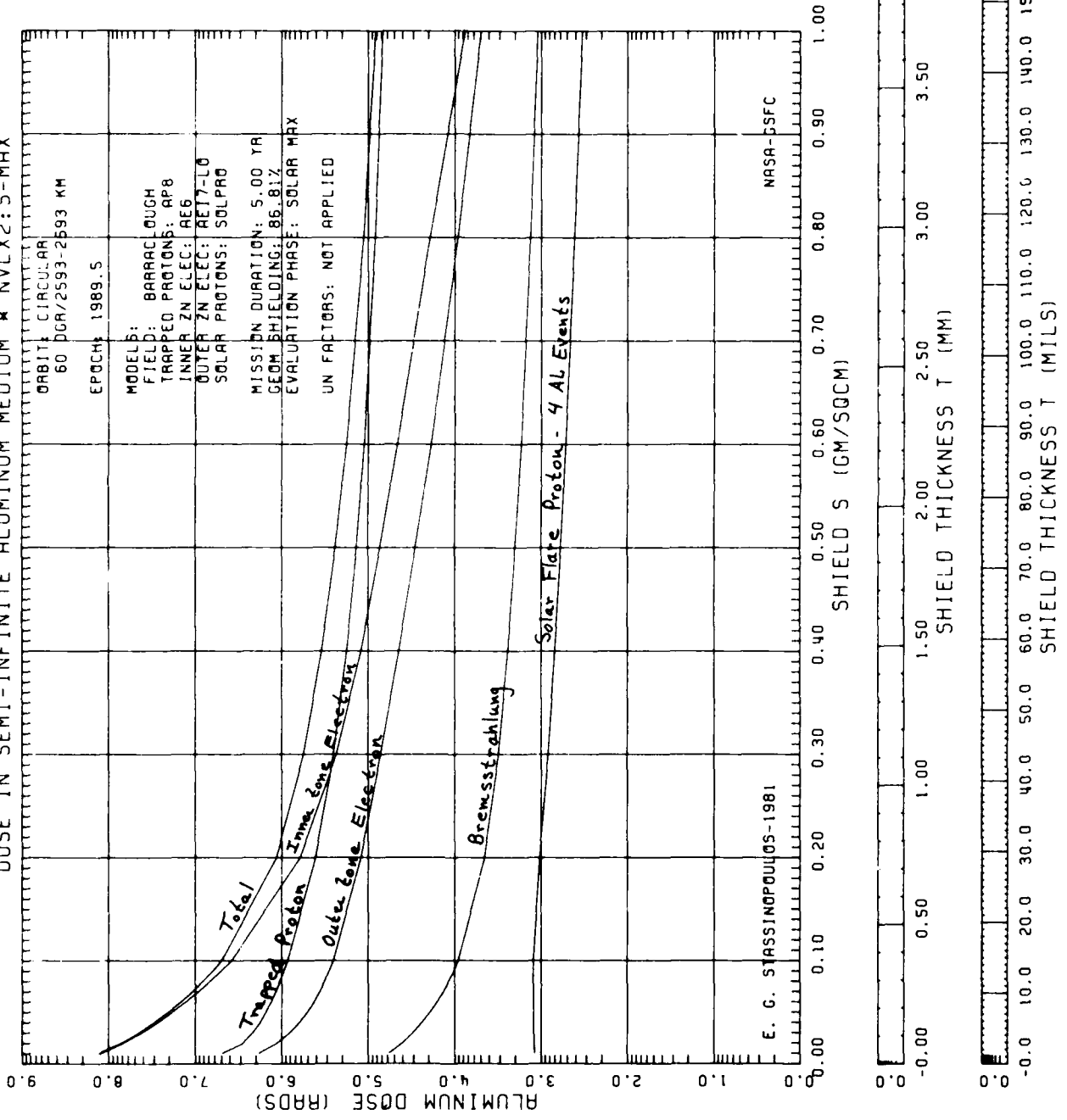
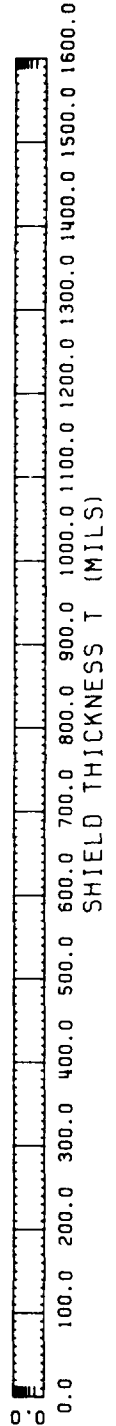
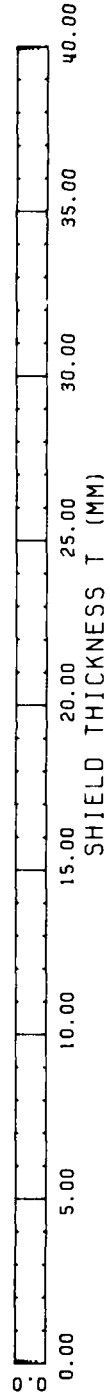
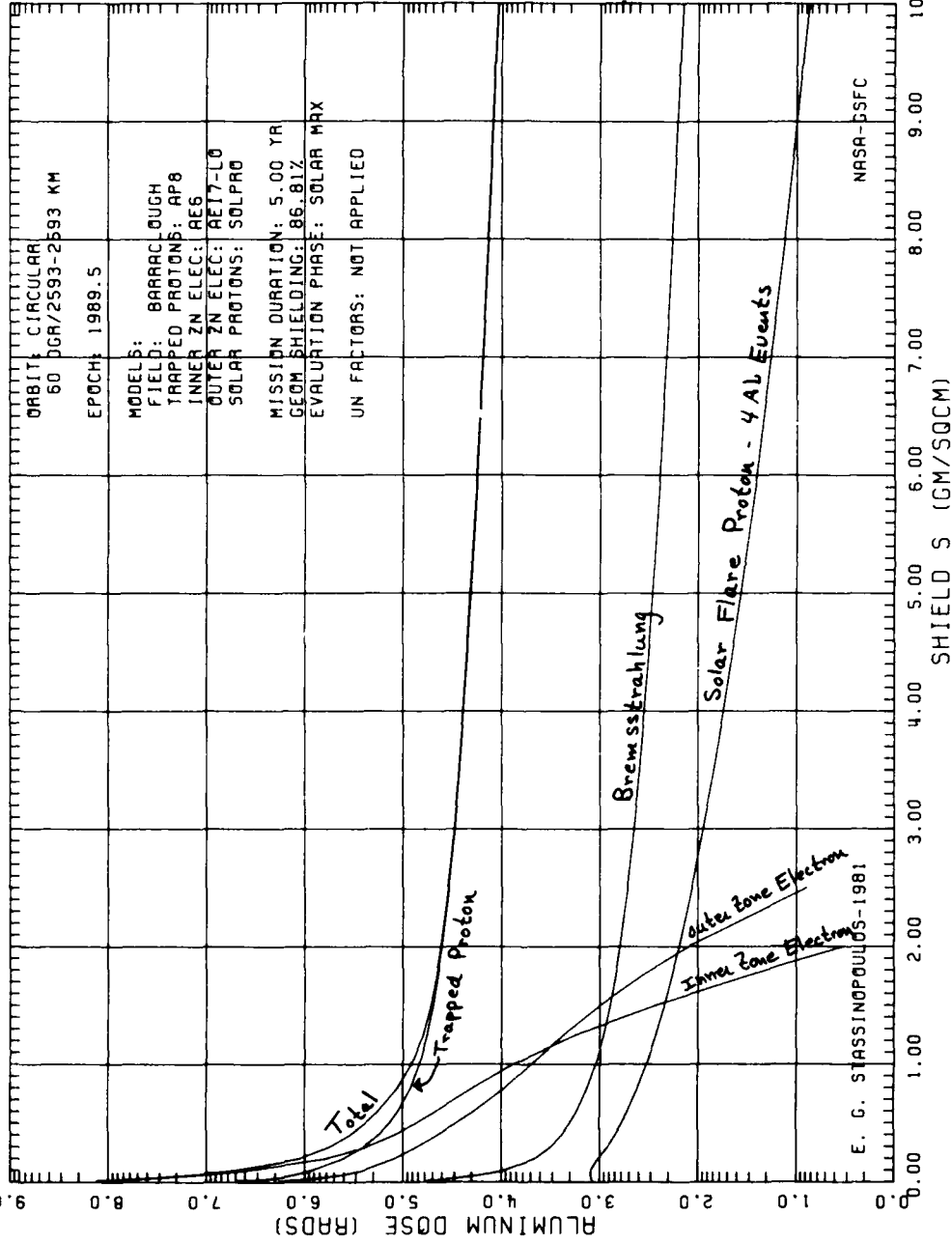




FIGURE 63

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX2:S-MAX



ORBIT: CIRCULAR  
60 DGR/2593-2593 KM  
EPOCH: 1989.5

MODELS:  
FIELD: BARRACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AET7-L0  
SOLAR PROTONS: SOLPRO

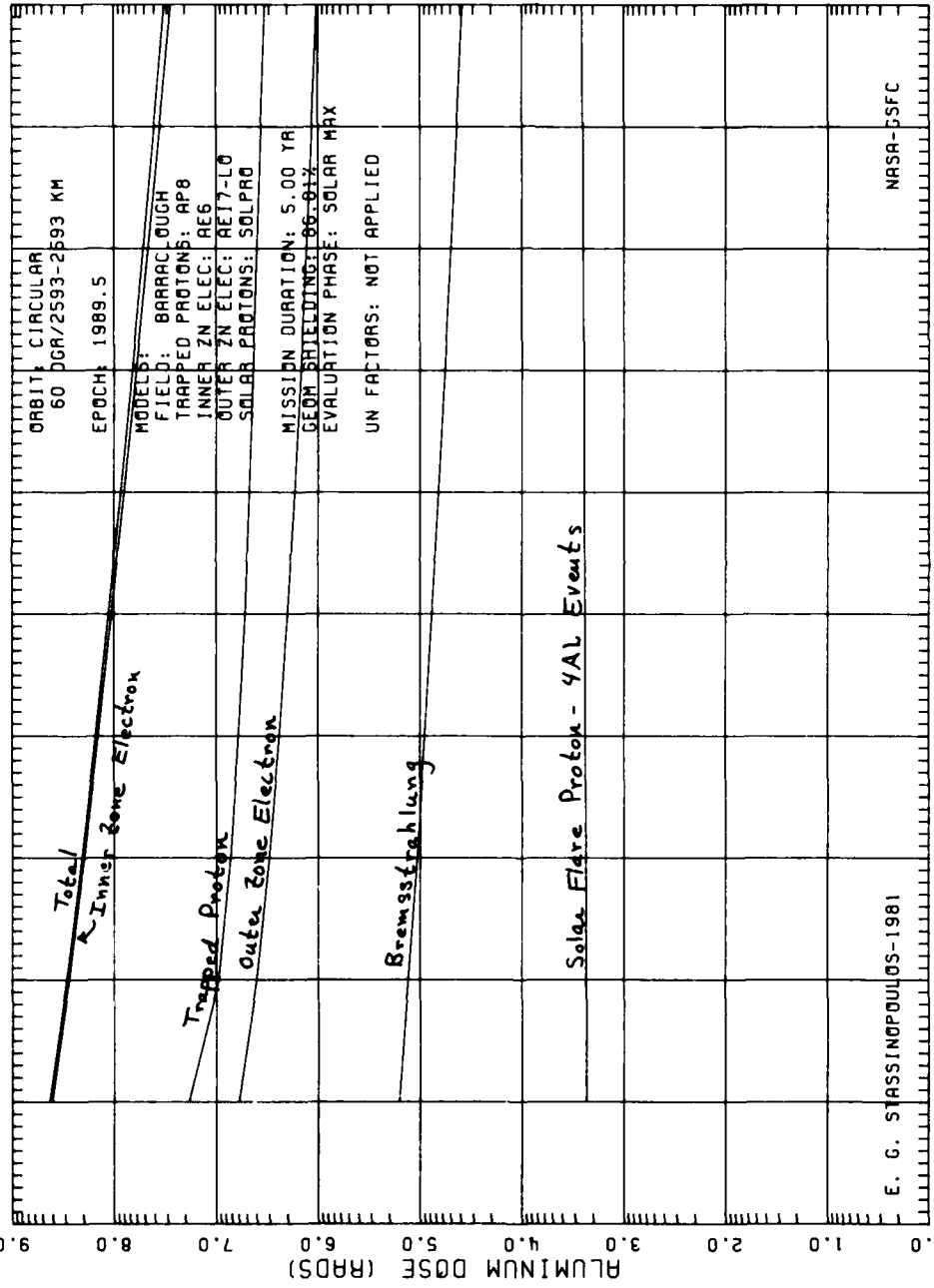
MISSION DURATION: 5.00 YR  
GEOM SHIELDING: 86.81%  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

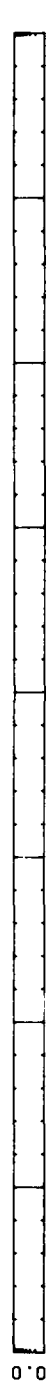
E. G. STASSINOPOULOS-1981

FIGURE 64

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX2:S-MAX



SHIELD S (GM/SQCM)



SHIELD THICKNESS T (MILS)

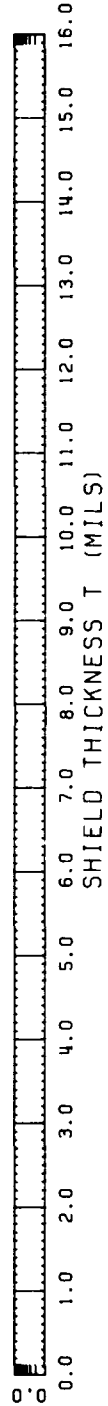


FIGURE 65

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX2:S-MAX

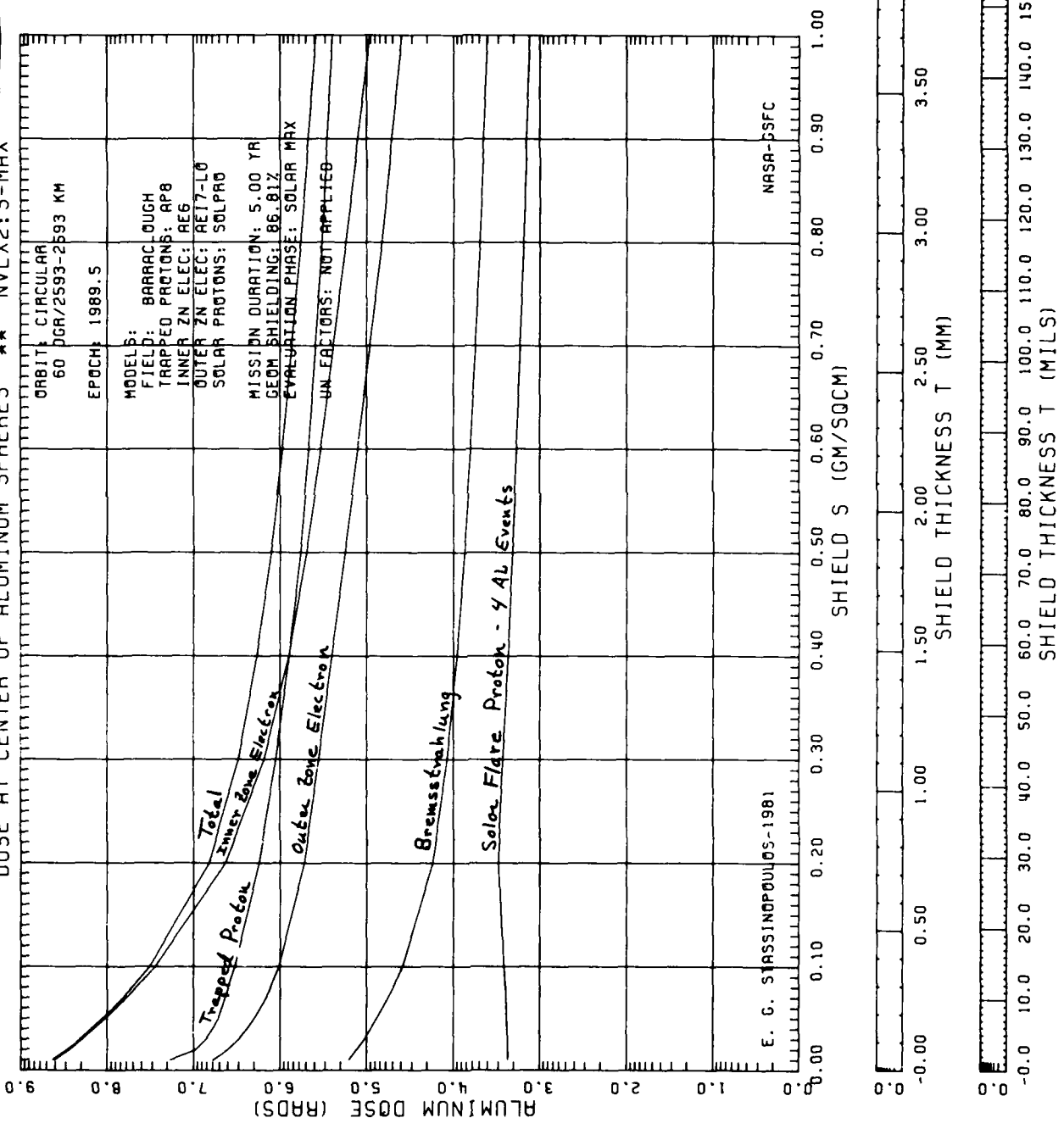
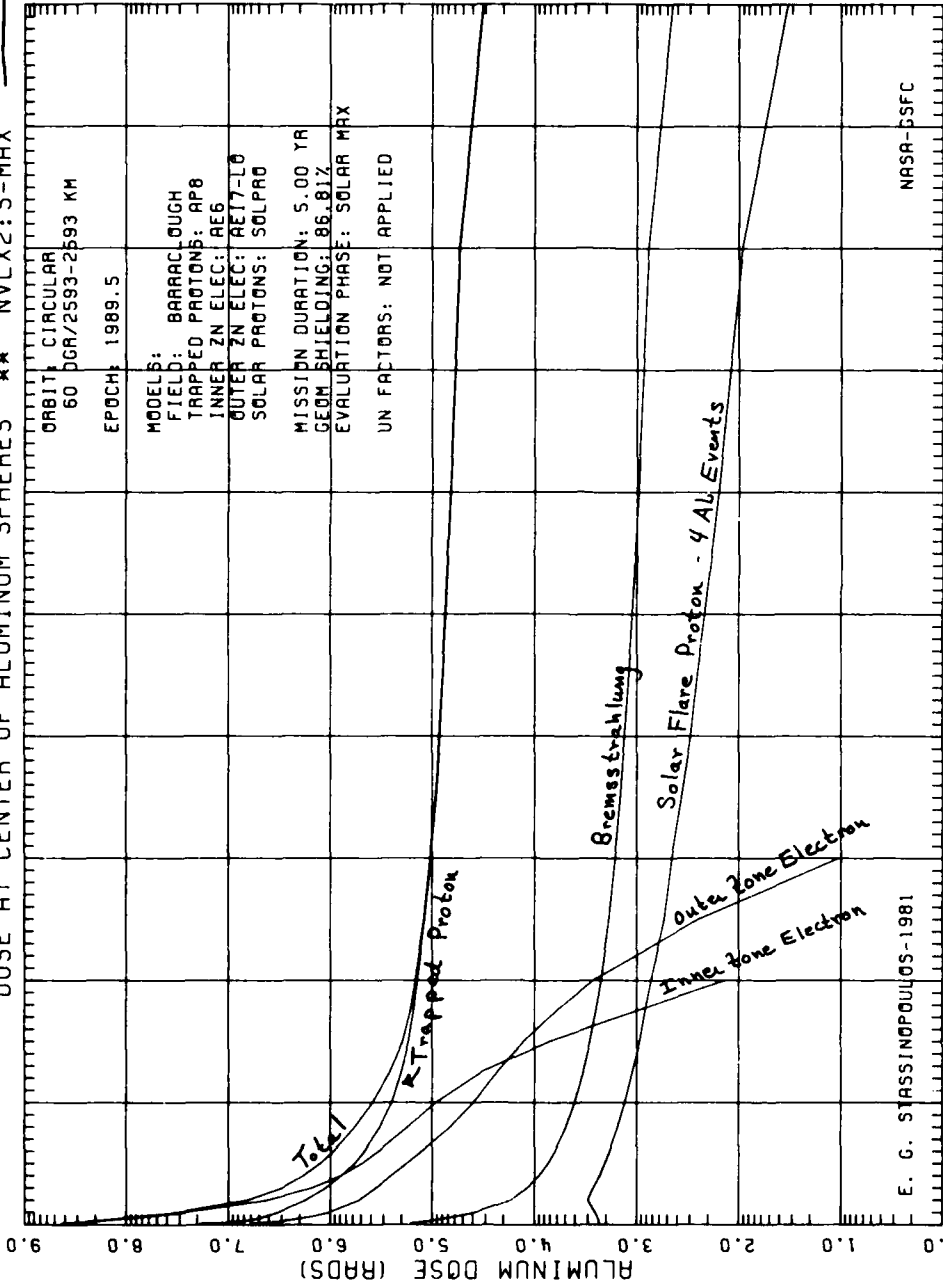


FIGURE 66

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX2:S-MAX



ORBIT: CIRCULAR  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: RET7-L0  
SOLAR PROTONS: SOLPRO

MISSION DURATION: 5.00 YR  
GEOM SHIELDING: 86.81%  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

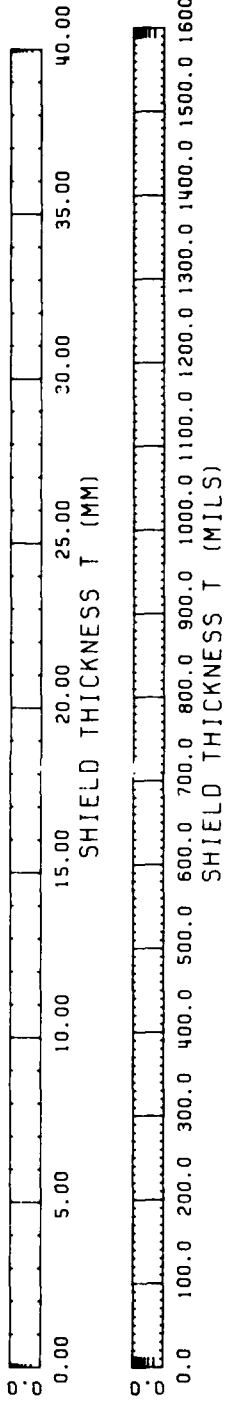
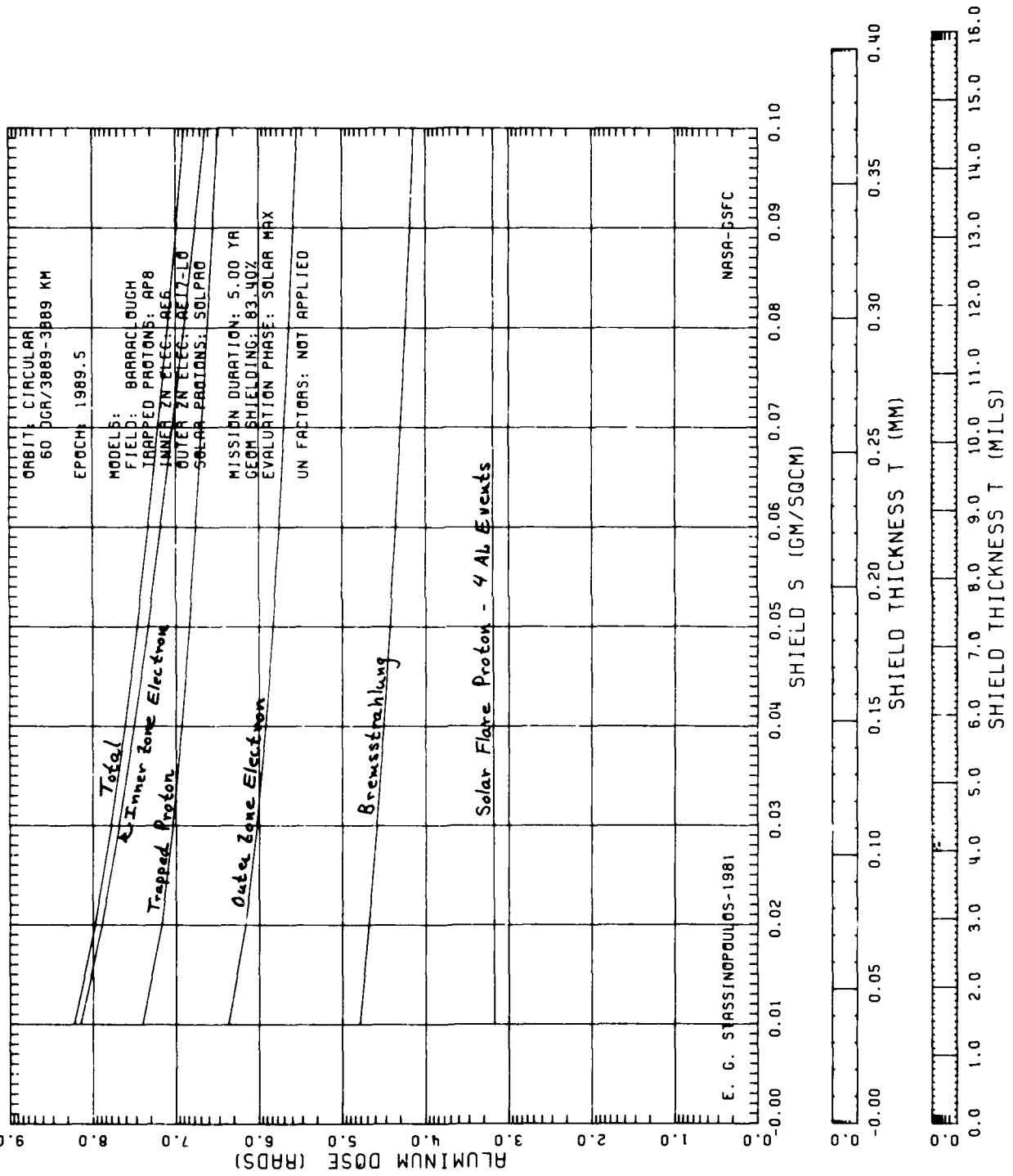


Figure 67

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX3: S-MAX



*Figure 68*  
DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX3: S-MAX

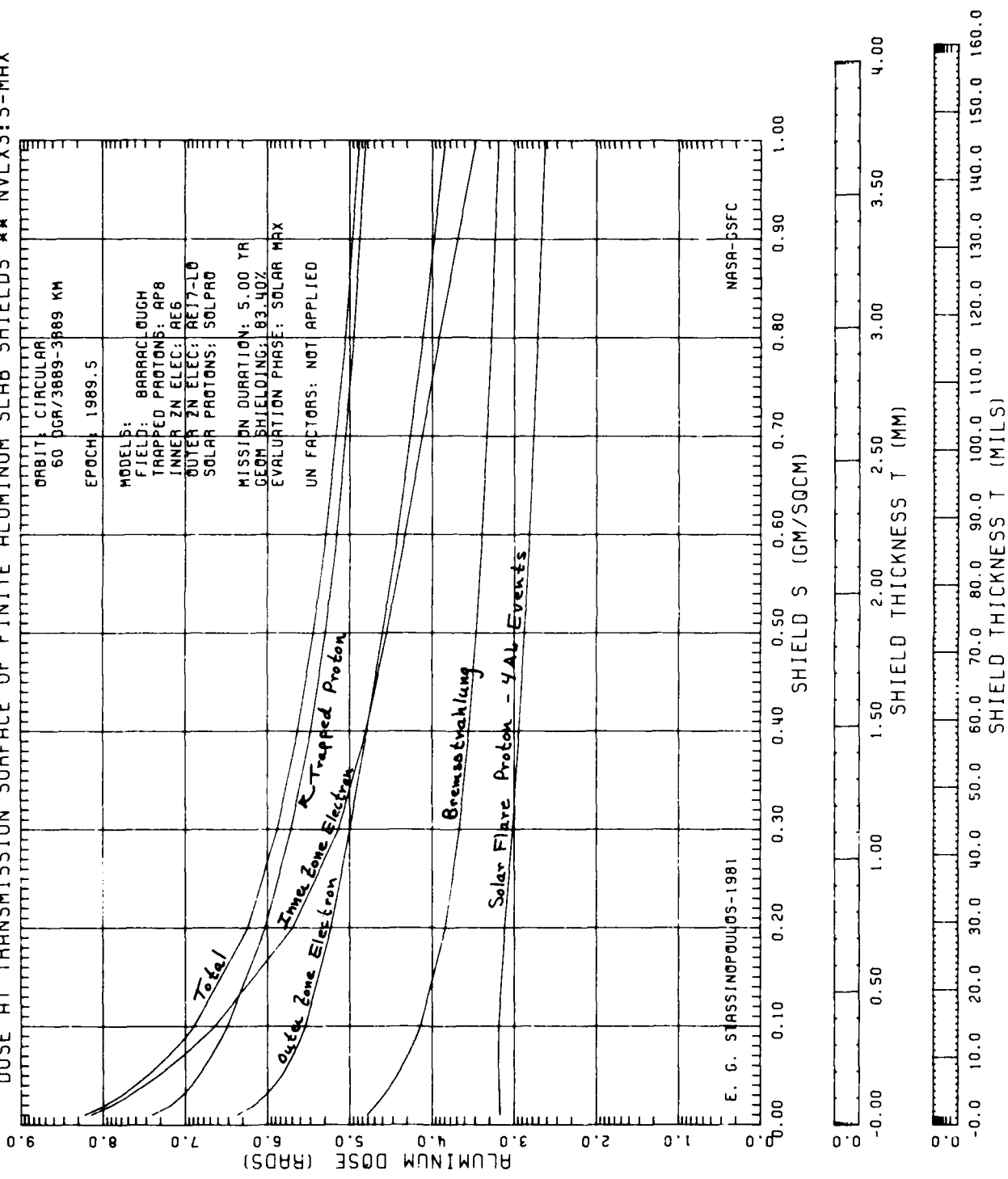
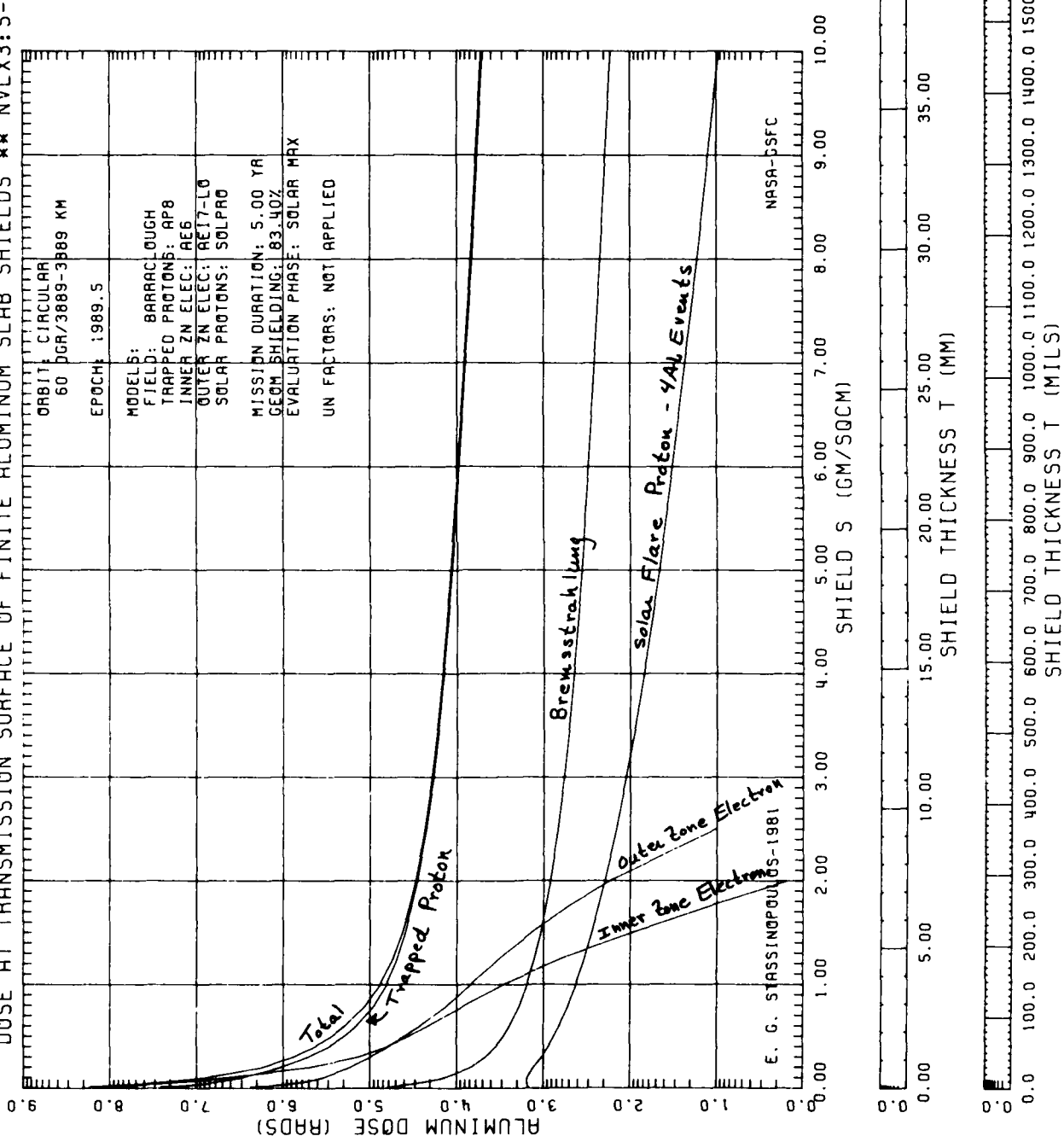


FIGURE 69

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX3:S-MAX



ORBIT: CIRCULAR  
60 DCR/3889-3889 KM  
EPOCH: 1989.5  
MODELS:  
FIELD: BARRACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AE17-L0  
SOLAR PROTONS: SOLPRO  
MISSION DURATION: 5.00 YR  
GEOM SHIELDING: 83.40%  
EVALUATION PHASE: SOLAR MAX  
UN FACTORS: NOT APPLIED

SHIELD S (GM/SQCM) 0.00 1.00 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00 10.00

SHIELD THICKNESS T (MM) 0.0 5.00 10.00 15.00 20.00 25.00 30.00 35.00 40.00

SHIELD THICKNESS T (MILS) 0.0 100.0 200.0 300.0 400.0 500.0 600.0 700.0 800.0 900.0 1000.0 1100.0 1200.0 1300.0 1400.0 1500.0 1600.0

Figure 70

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX3:S-MAX

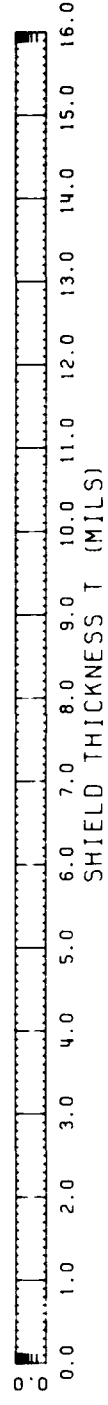
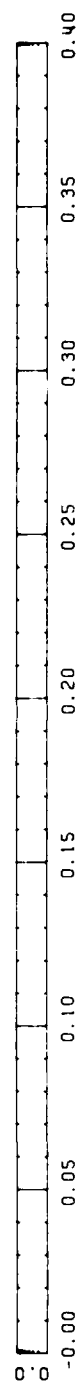
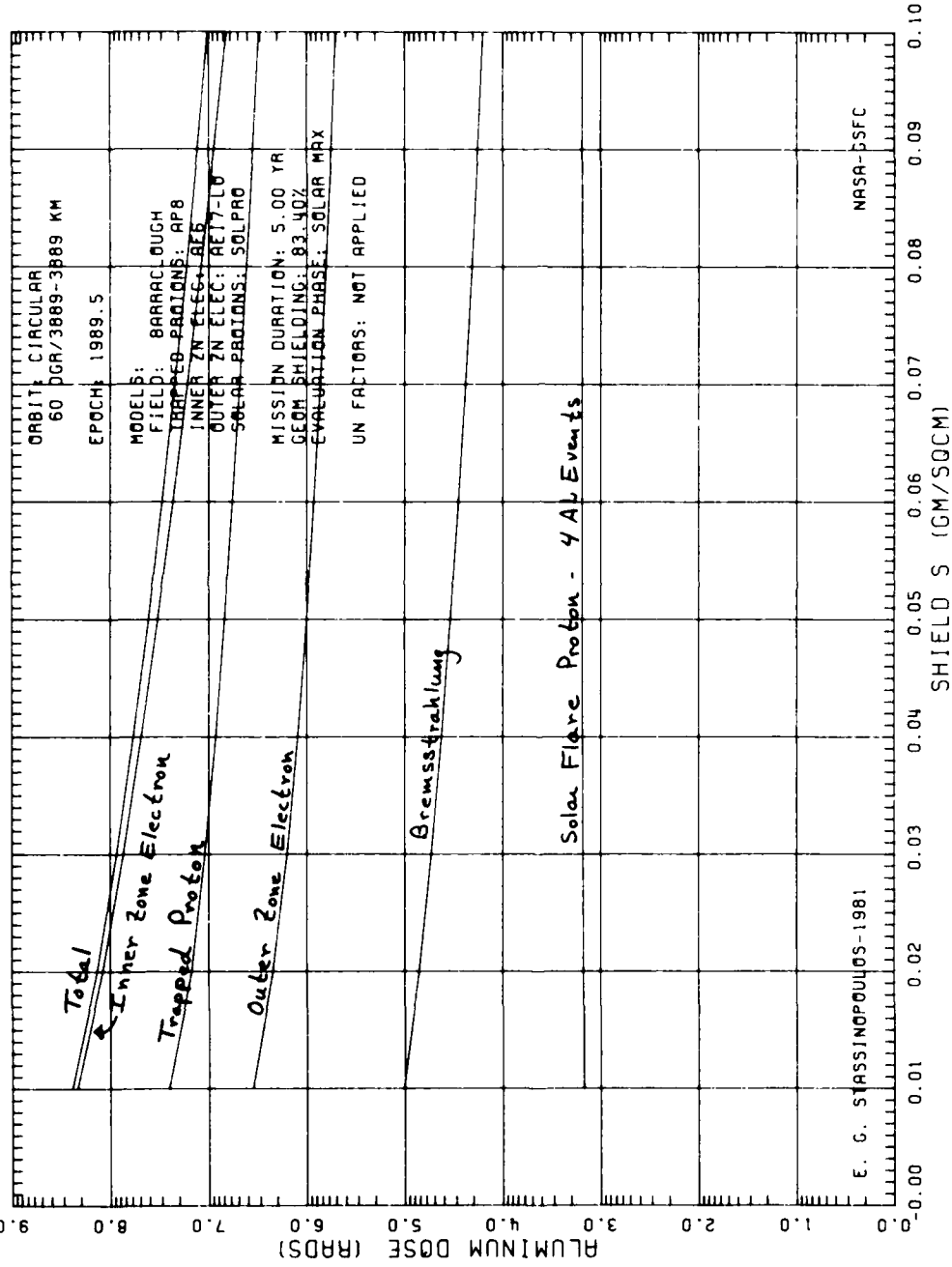




Figure 71

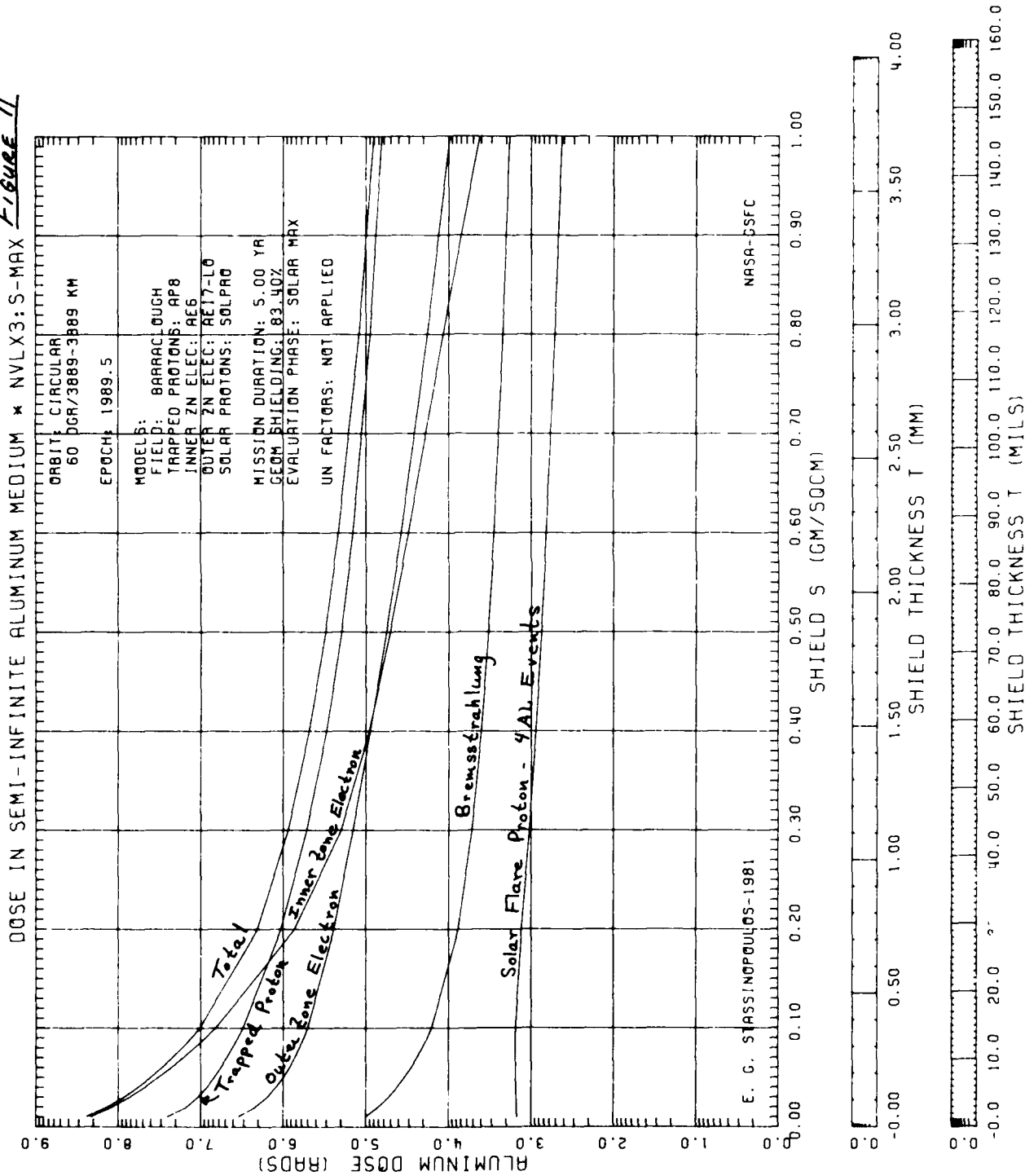


FIGURE 72

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX3: S-MAX

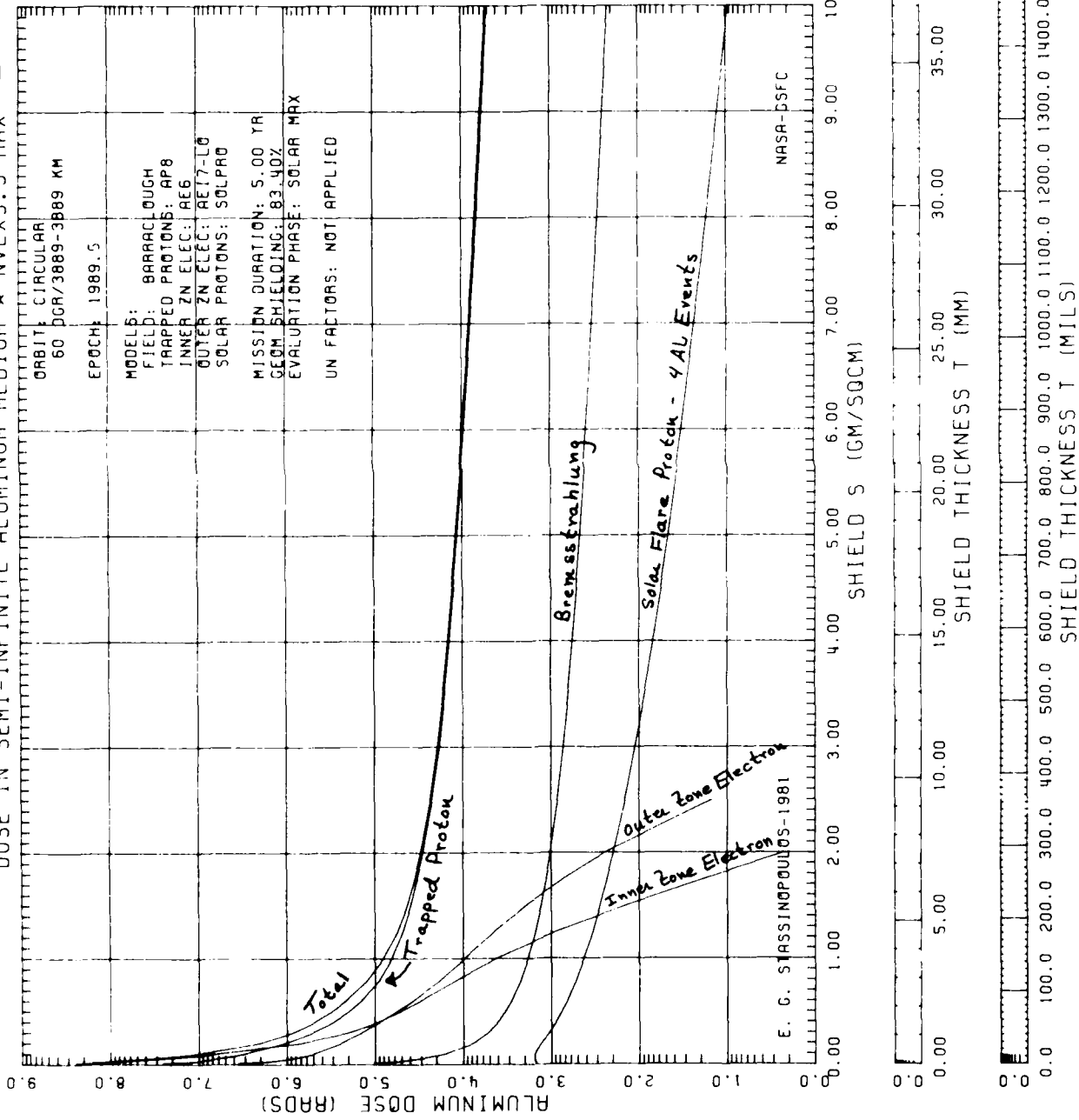
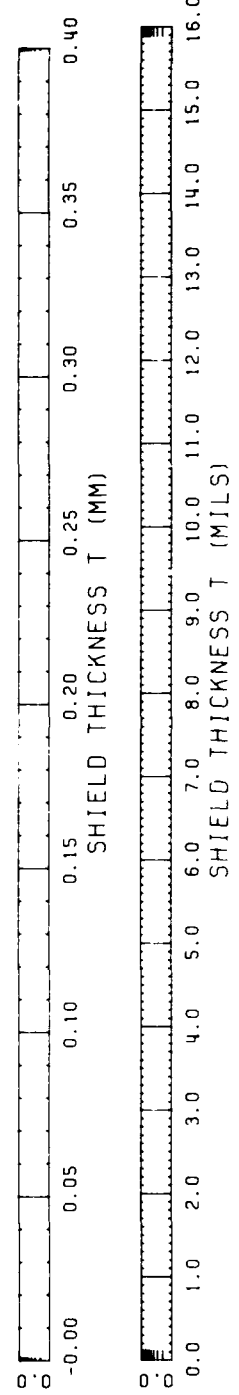
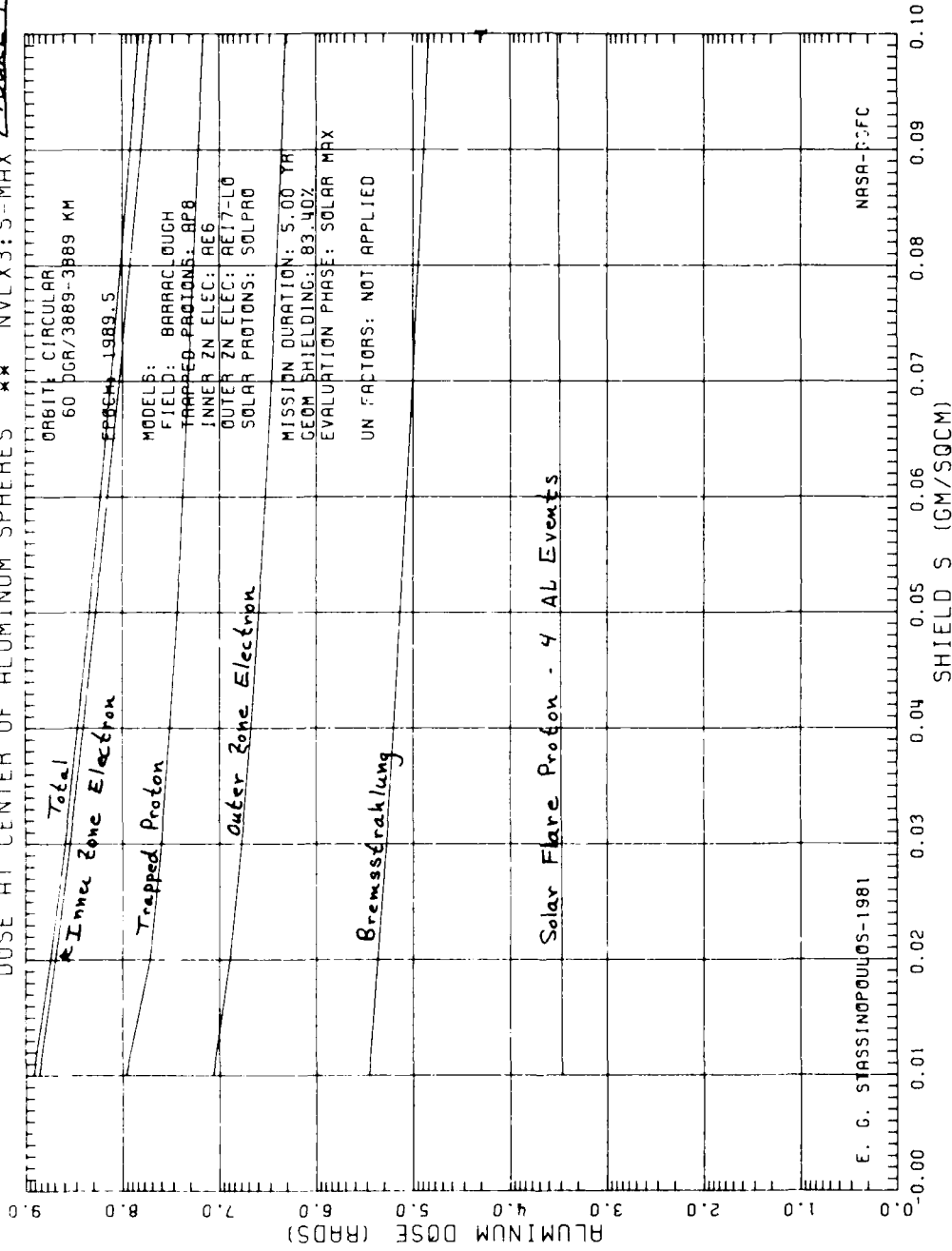


FIGURE 73

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX3: S-MAX



E. G. STASSINPOULOS-1981

NASA-SSFC

FIGURE 74

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX3: S-MAX

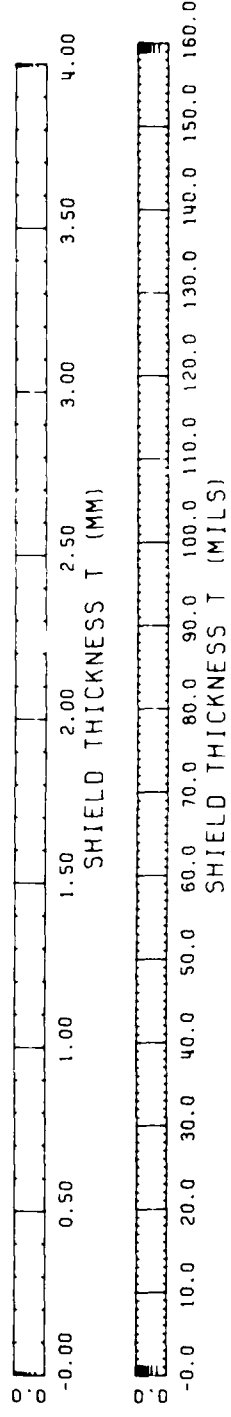
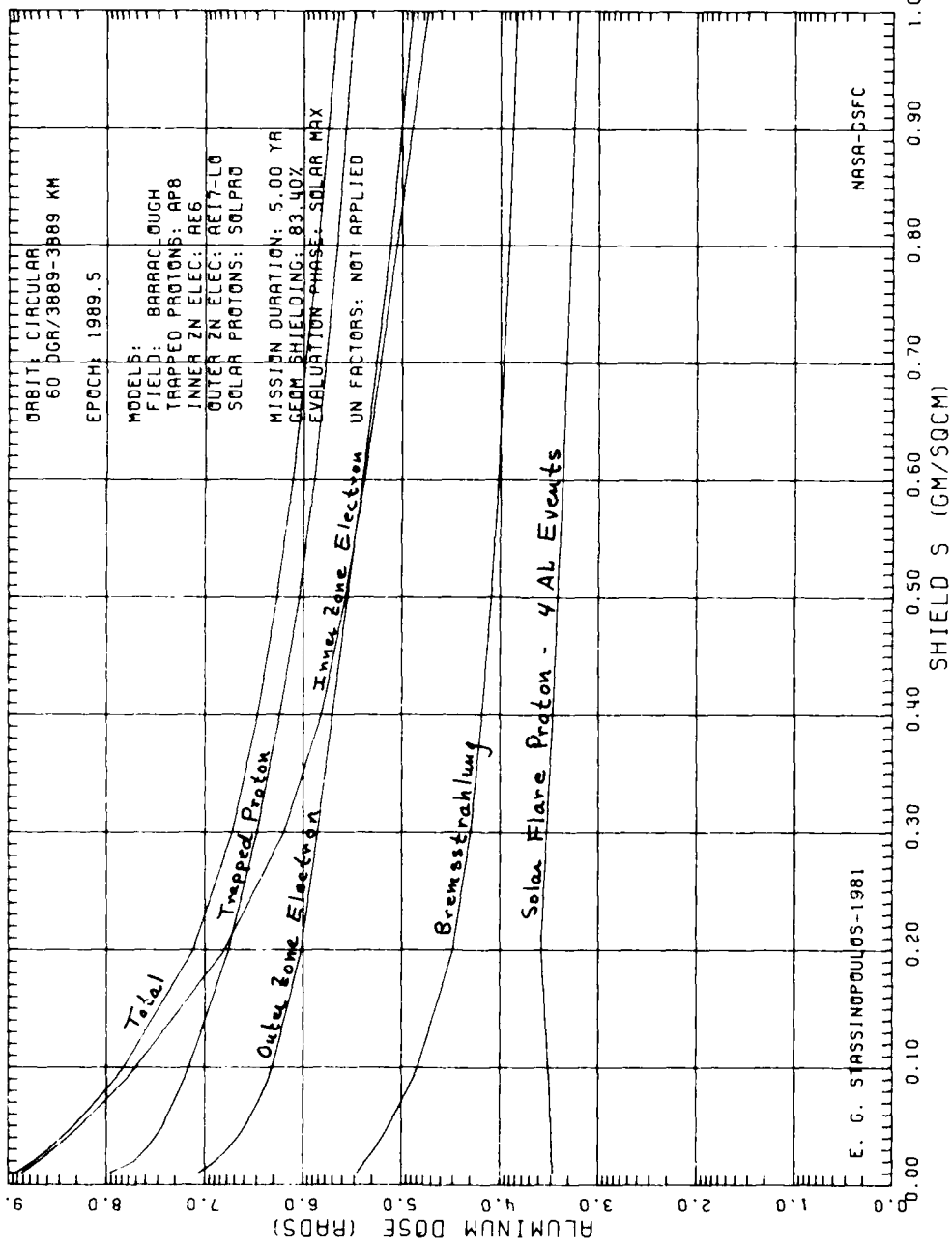
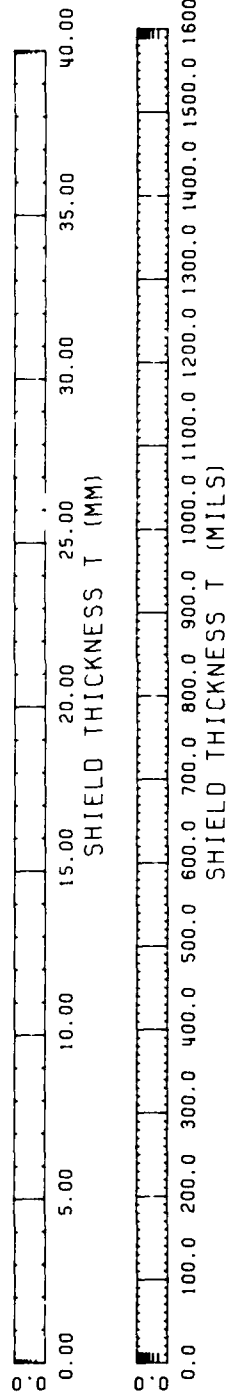
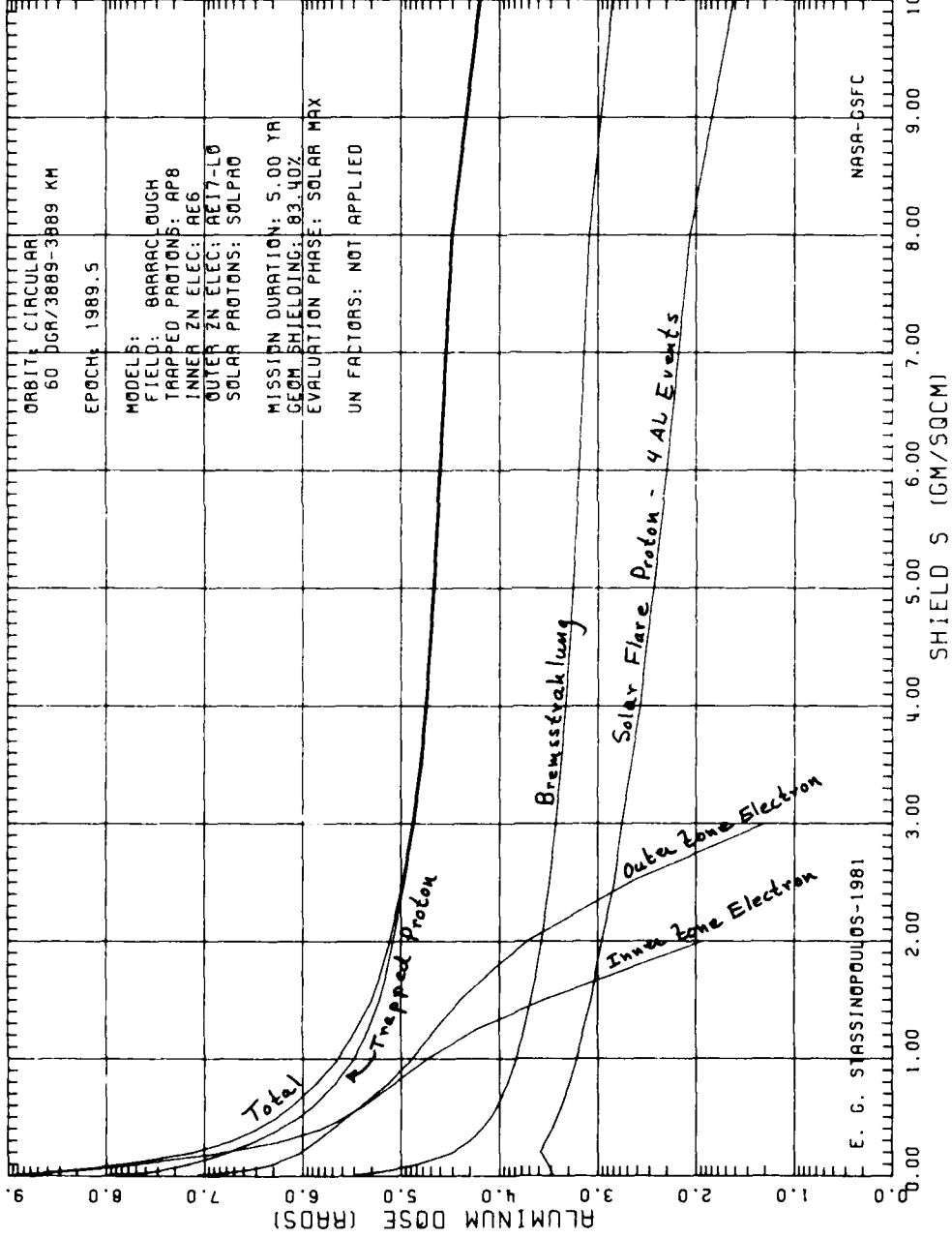


FIGURE 75

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX3: S-MAX



ORBIT: CIRCULAR  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODEL: BARRACLOUGH  
FIELD: TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: ACT7-L0  
SOLAR PROTONS: SOLPAD

MISSION DURATION: 5.00 YR  
GEOM SHIELDING: 83.40Z  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

E. G. STASSINopoulos-1981

FIGURE 76

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX4:S-MAX

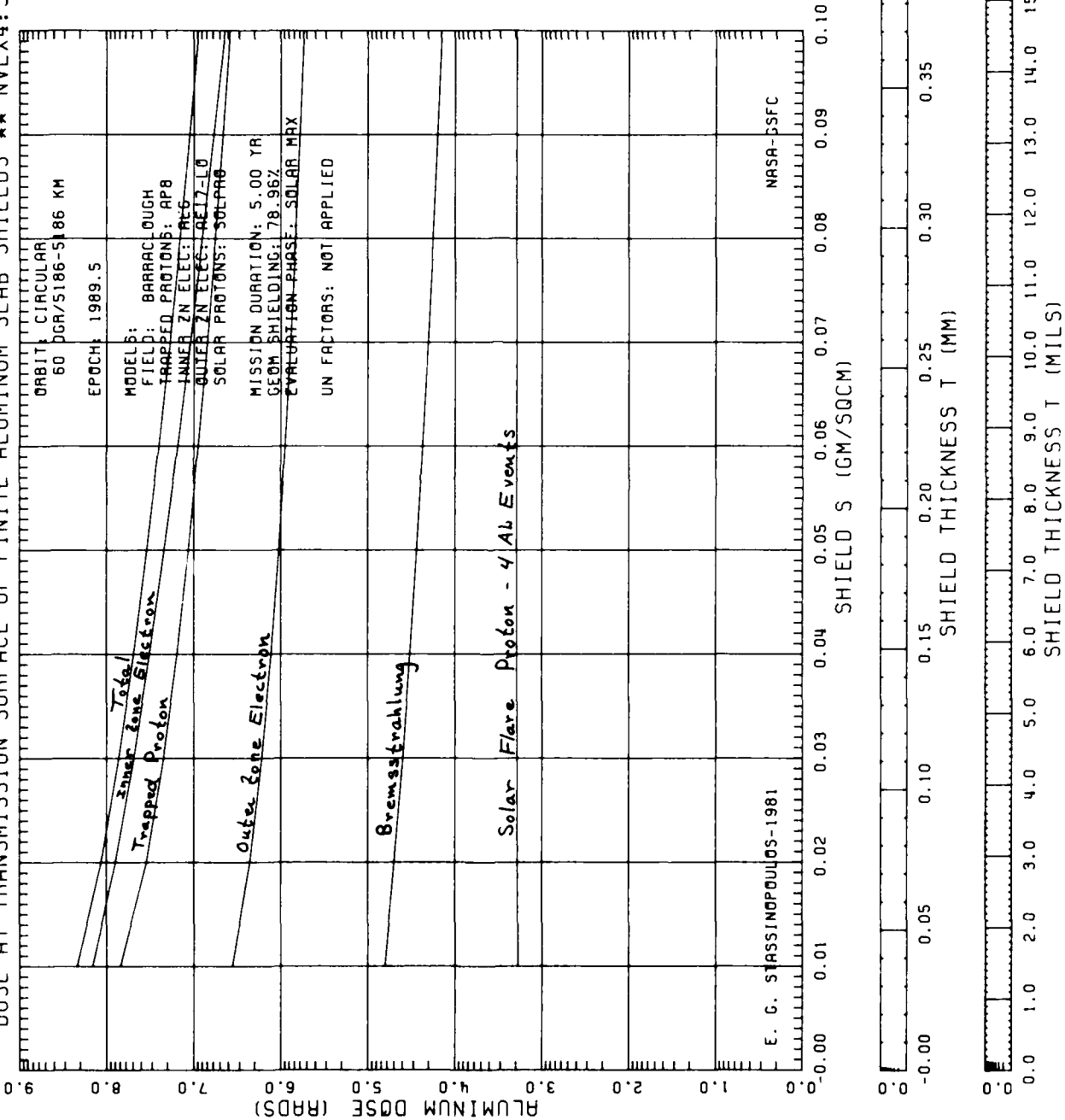


FIGURE 77

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX4: S-MAX

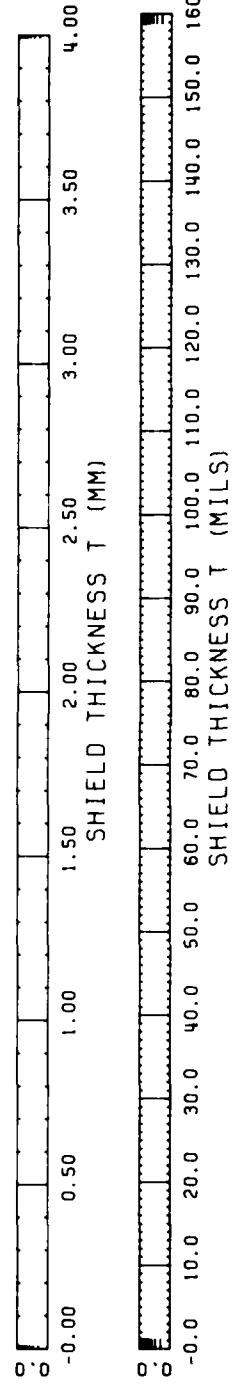
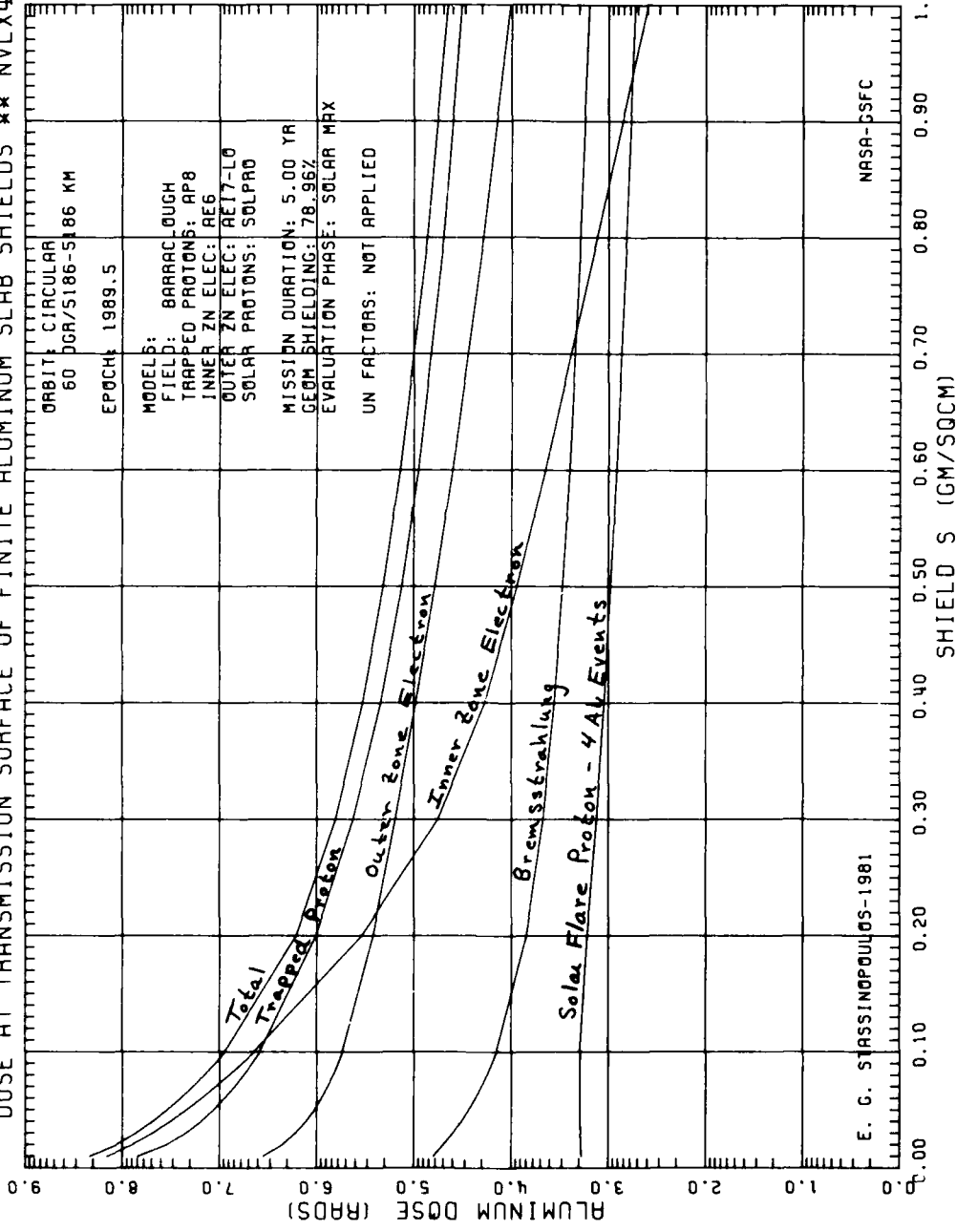


Figure 78

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX4: S-MAX

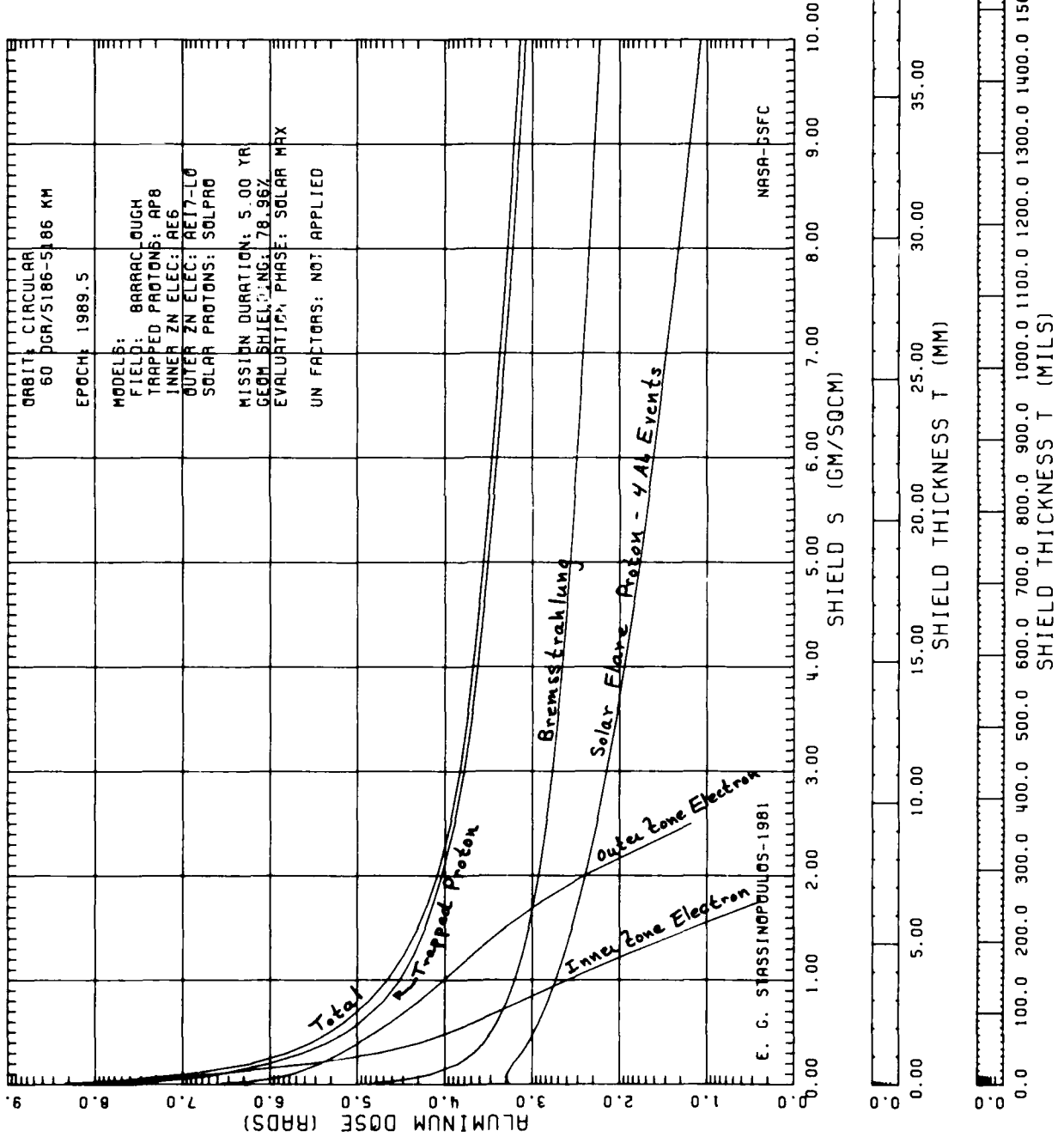




FIGURE 79

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX4:S-MAX

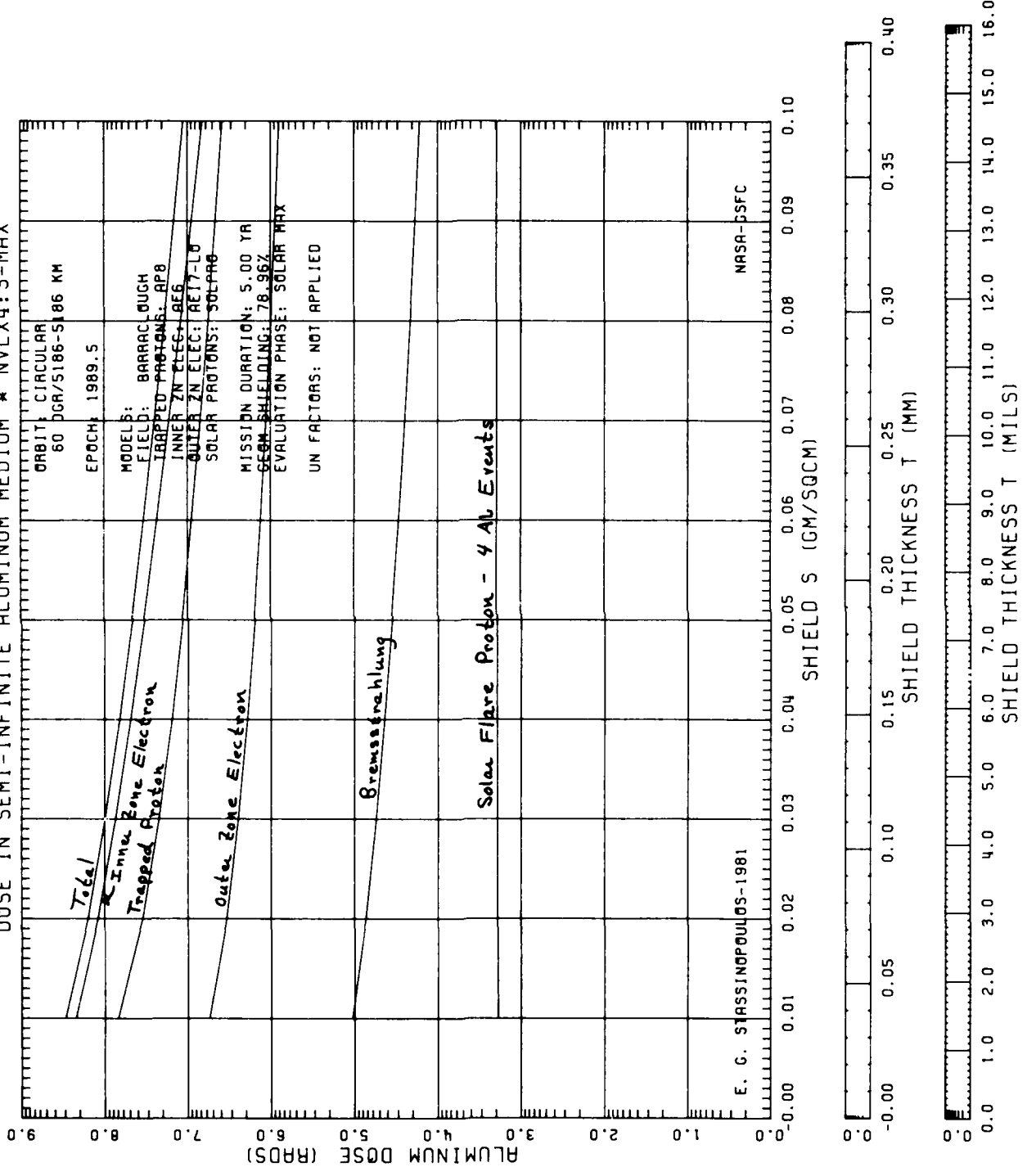


Figure 80

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVL4:S-MAX

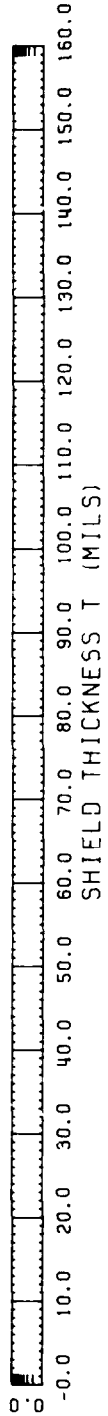
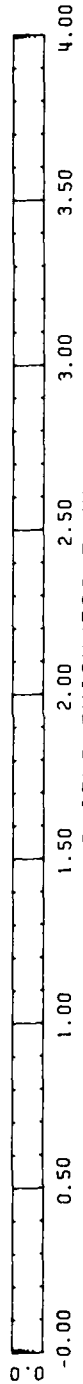
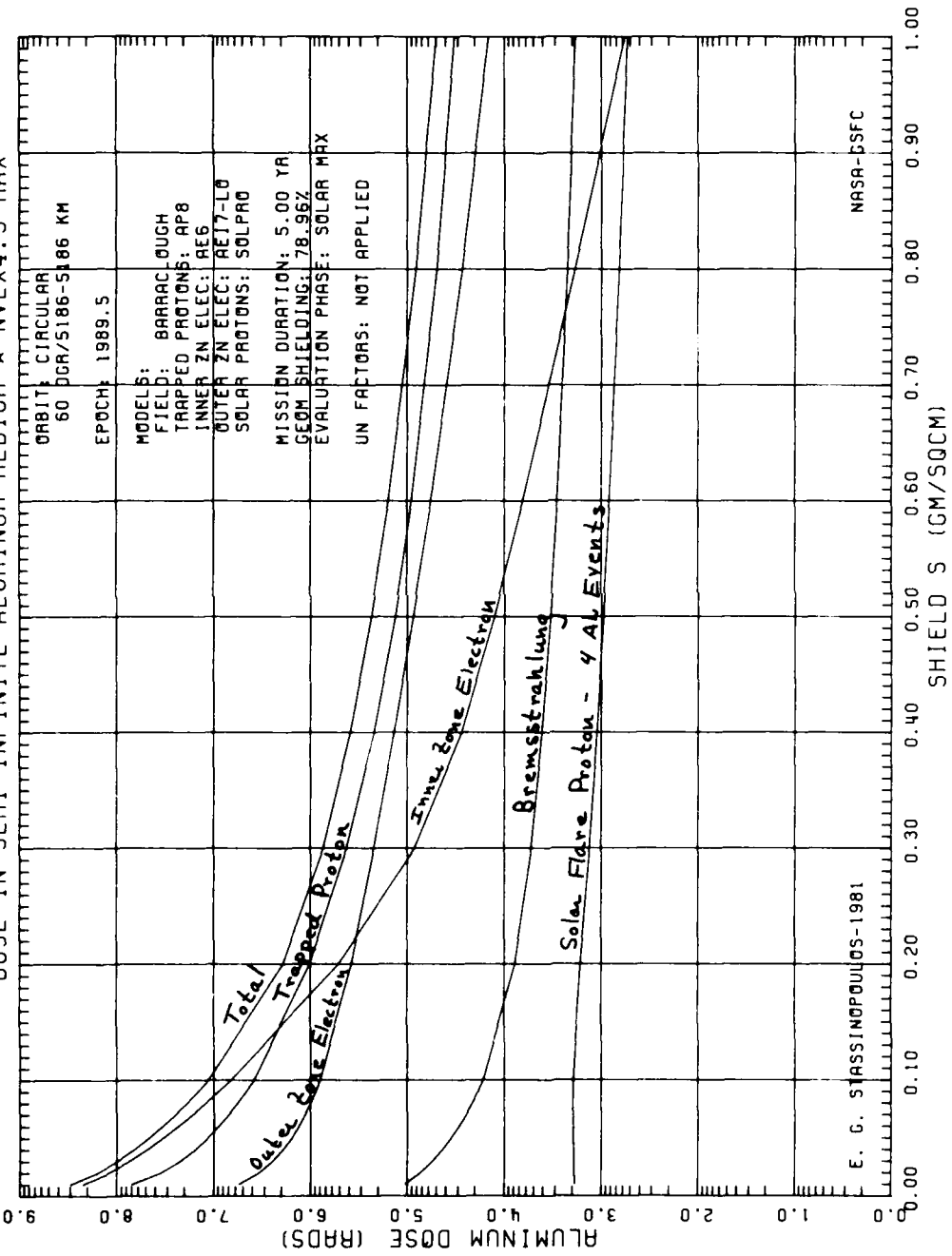
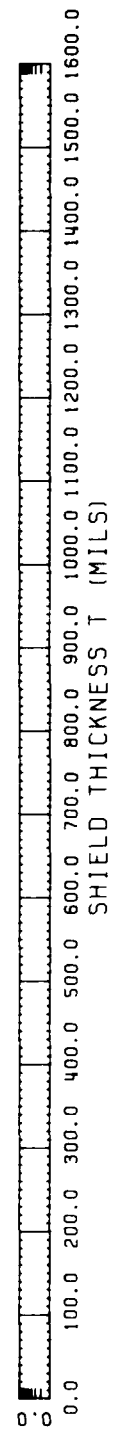
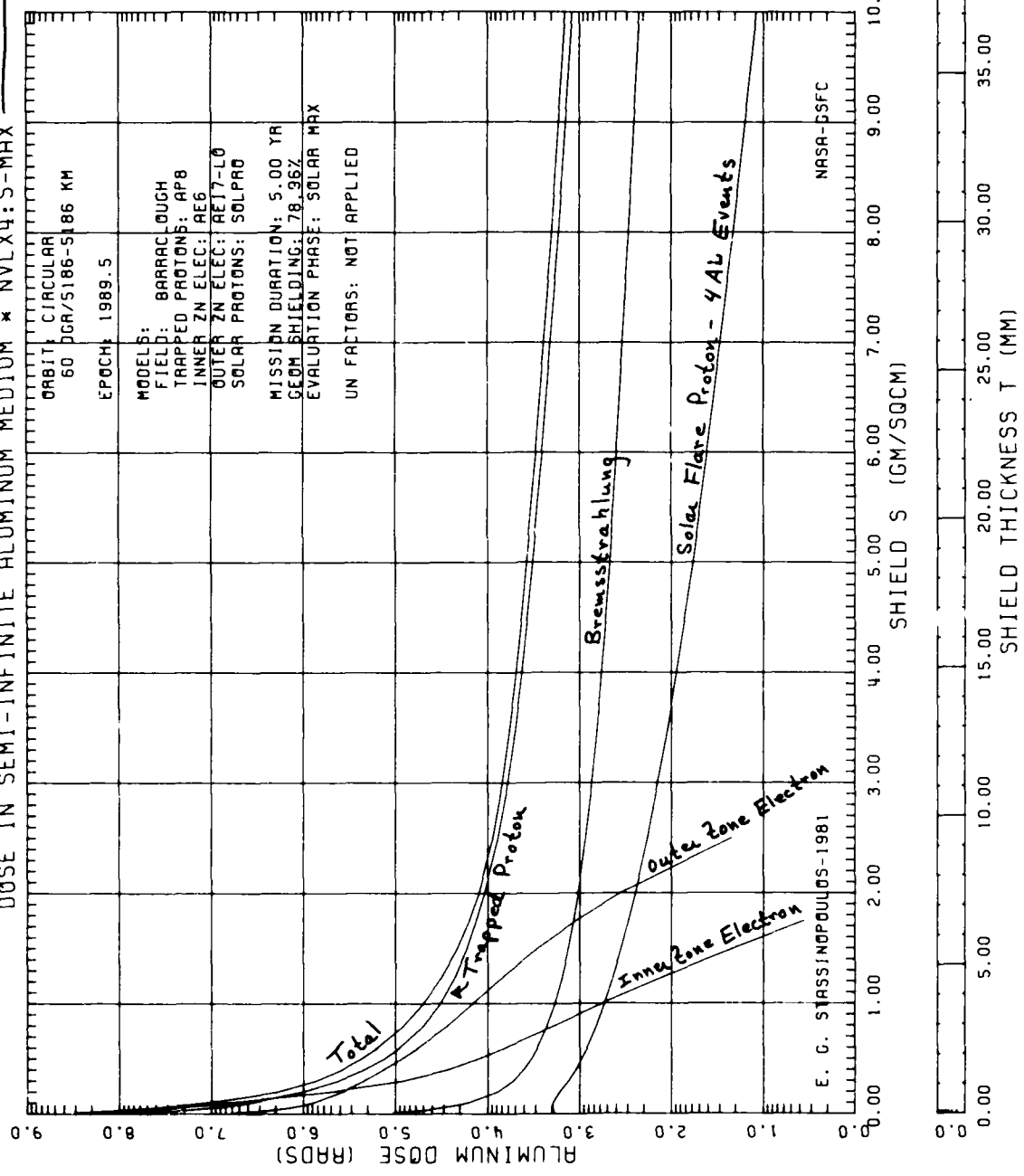


FIGURE 81

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX4: S-MAX



E. G. STASSINPOULOS-1981

FIGURE 82

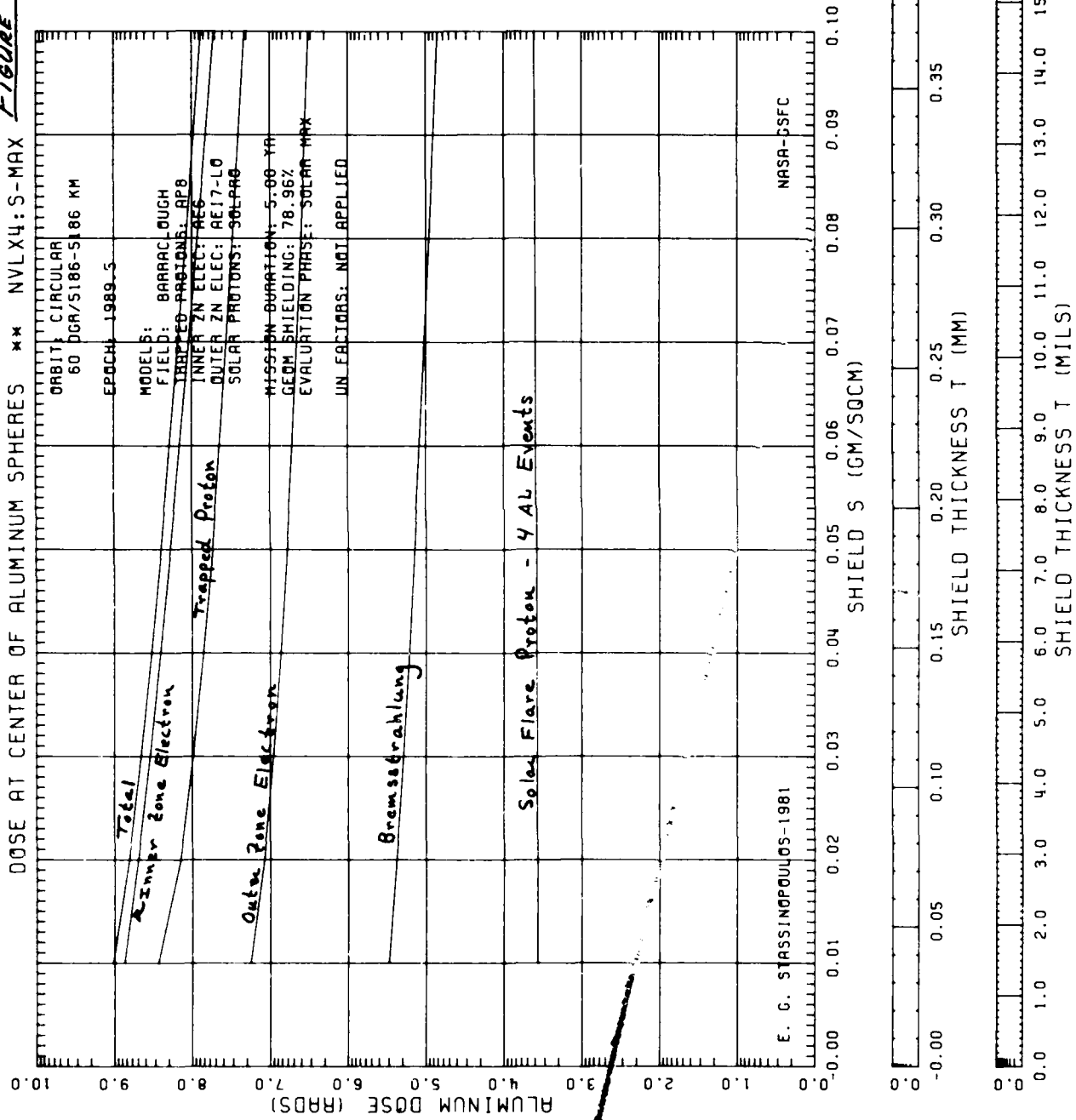
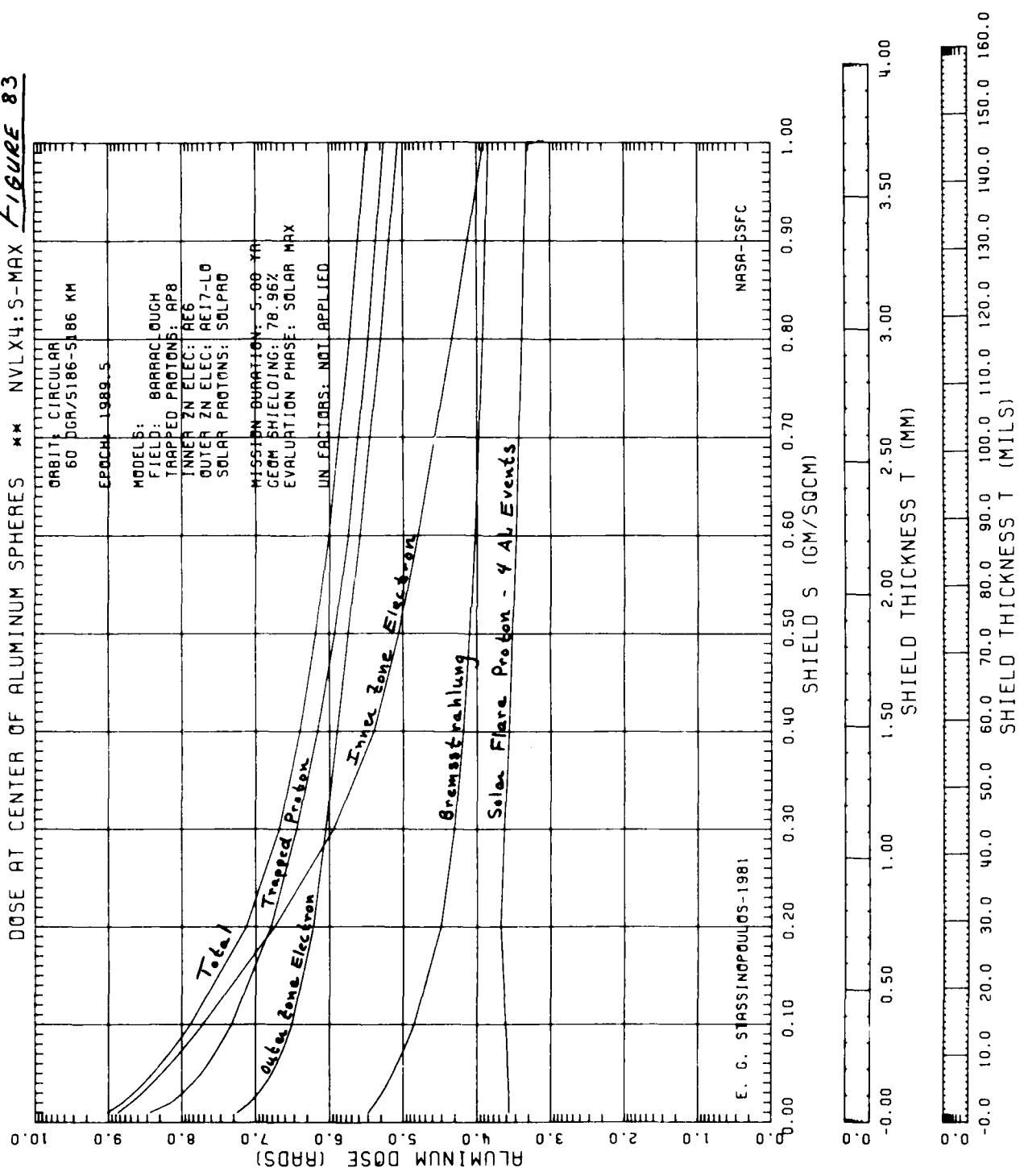


FIGURE 83



DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX4: S-MAX *FIGURE 84*

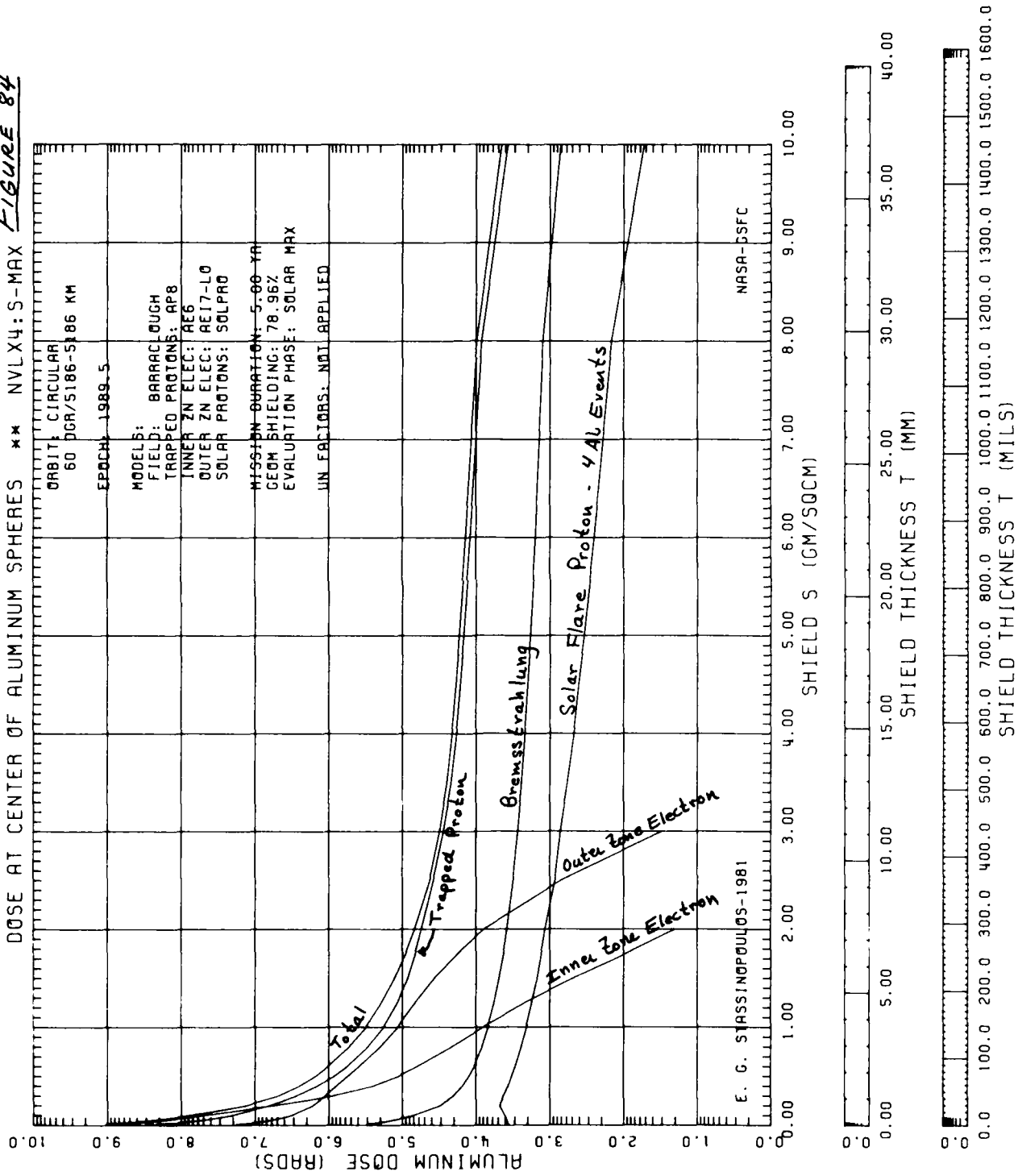
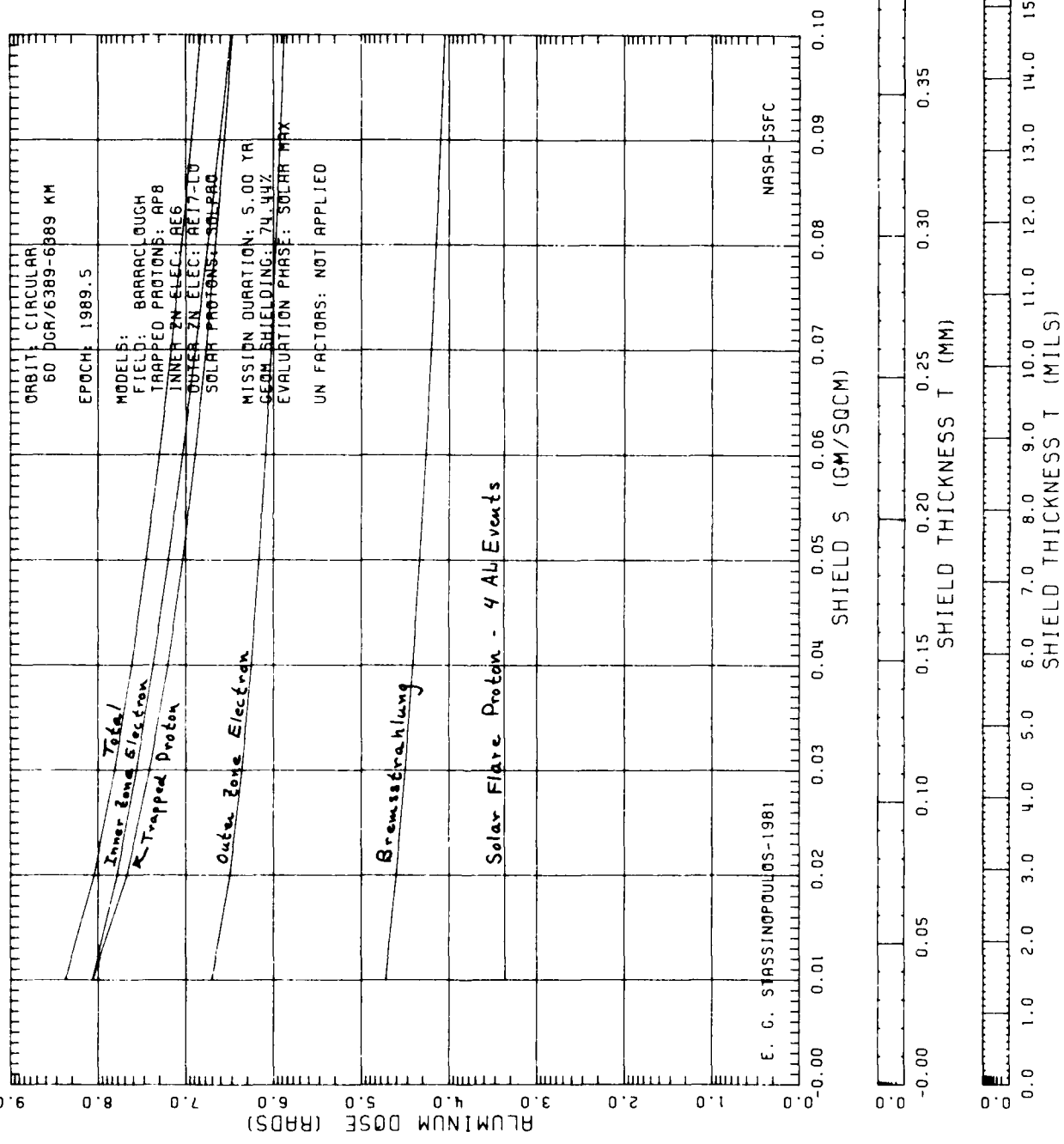


FIGURE 85

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX5:S-MAX



**FIGURE 86**  
 \*\* NVLX5: S-MAX

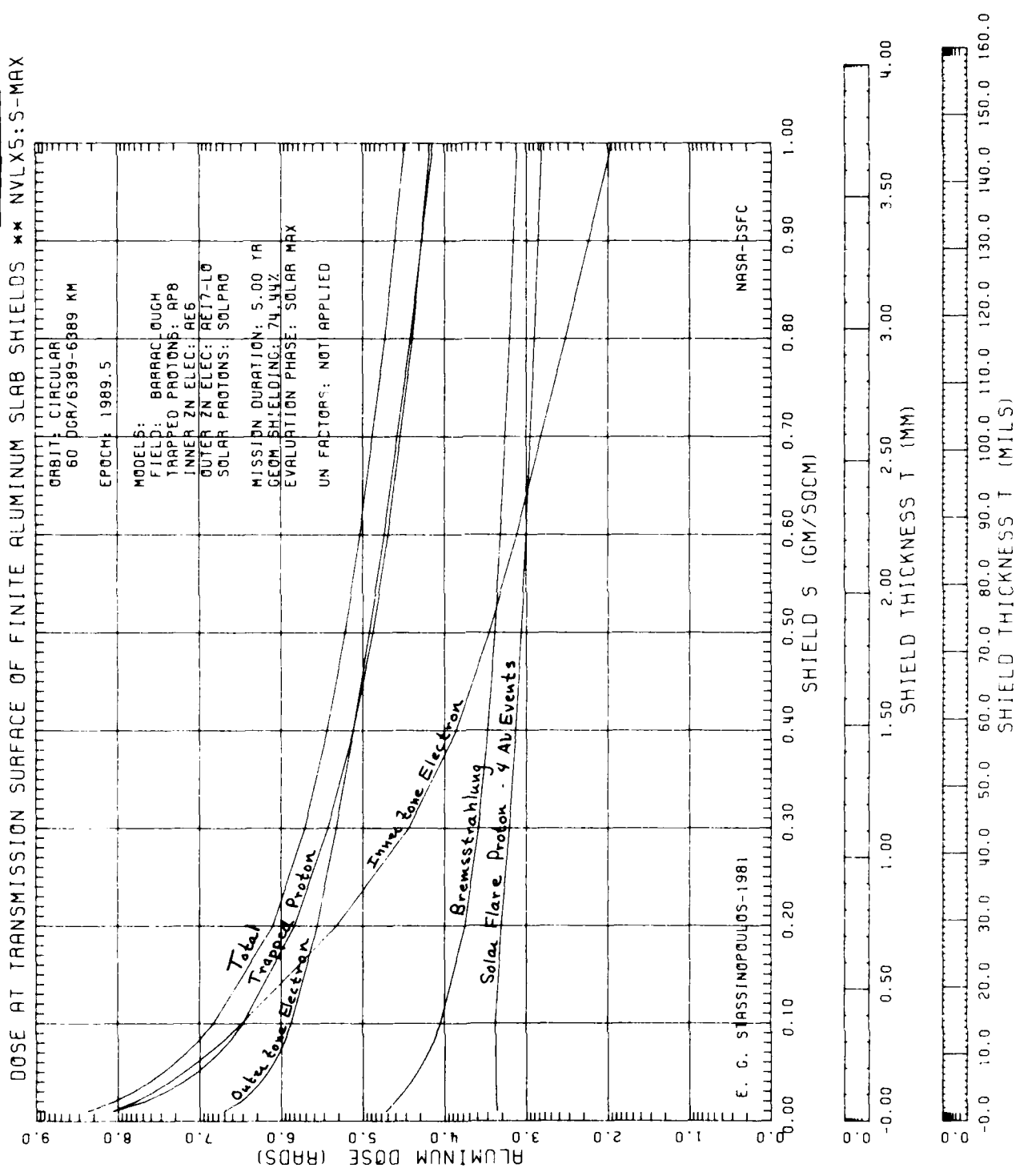
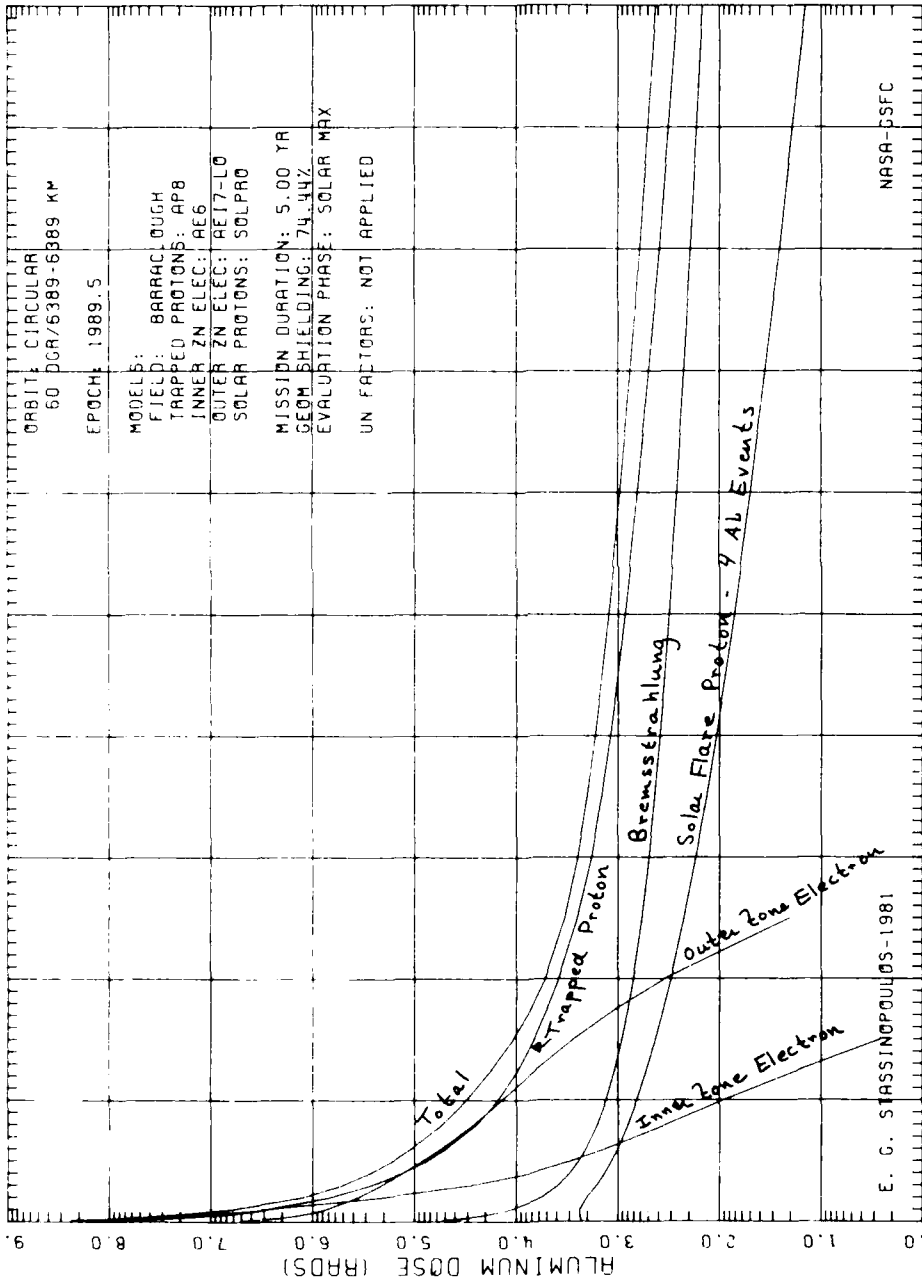


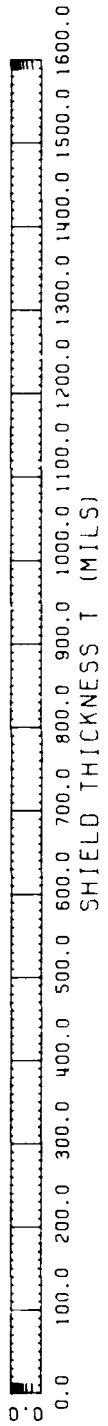
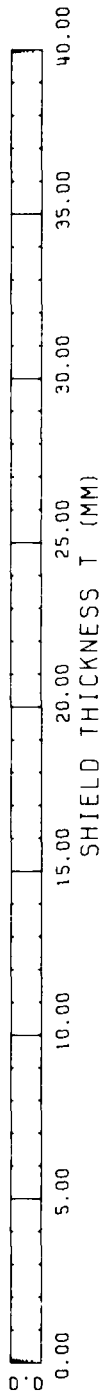


FIGURE 87

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX5: S-MAX



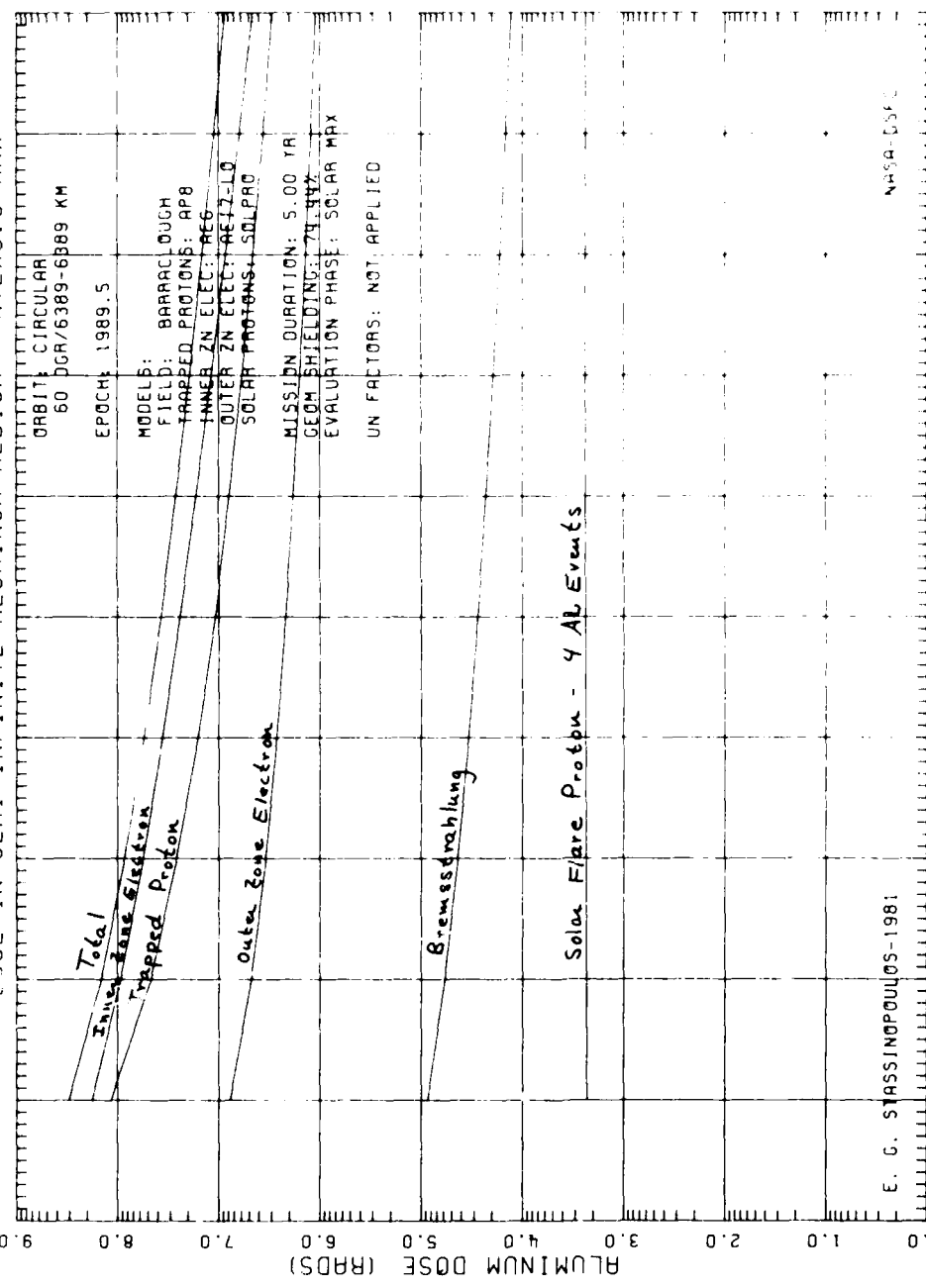
60 DCA/6389-6889 KM  
EPOCH: 1989.5  
MODELS:  
FIELD: BARRACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE7-LO  
SOLAR PROTONS: SOLPRO  
MISSION DURATION: 5.00 YR  
GEOM SHIELDING: 74.44%  
EVALUATION PHASE: SOLAR MAX  
UN FACTORS: NOT APPLIED



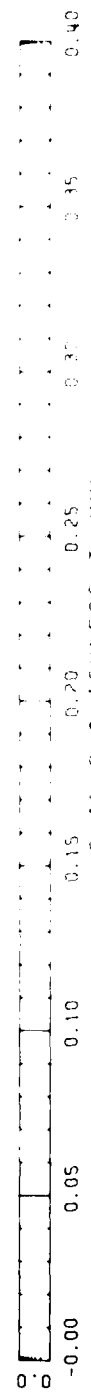
E. G. STASINAKOPOULOS-1981

FIGURE 88

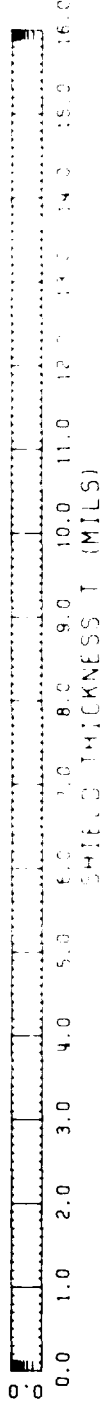
DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX5:S-MAX



SHIELD S (CM/SOCCM)



SHIELD THICKNESS T (MILS)



E. G. STASSINOPoulos-1981

NASA-656C

FIGURE 89

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX:S:5 MAX

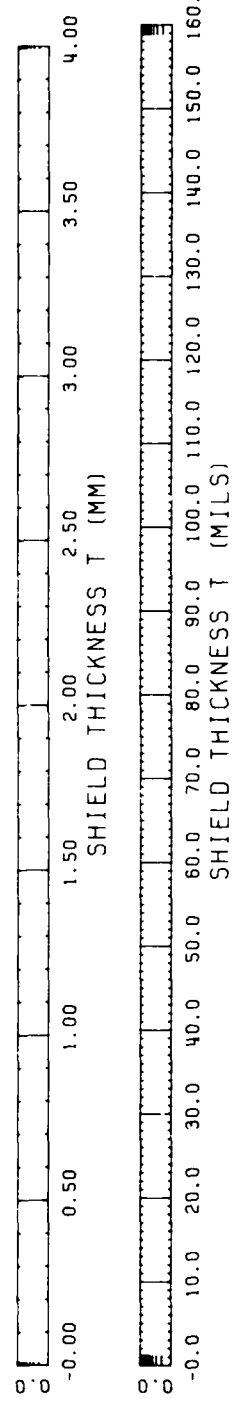
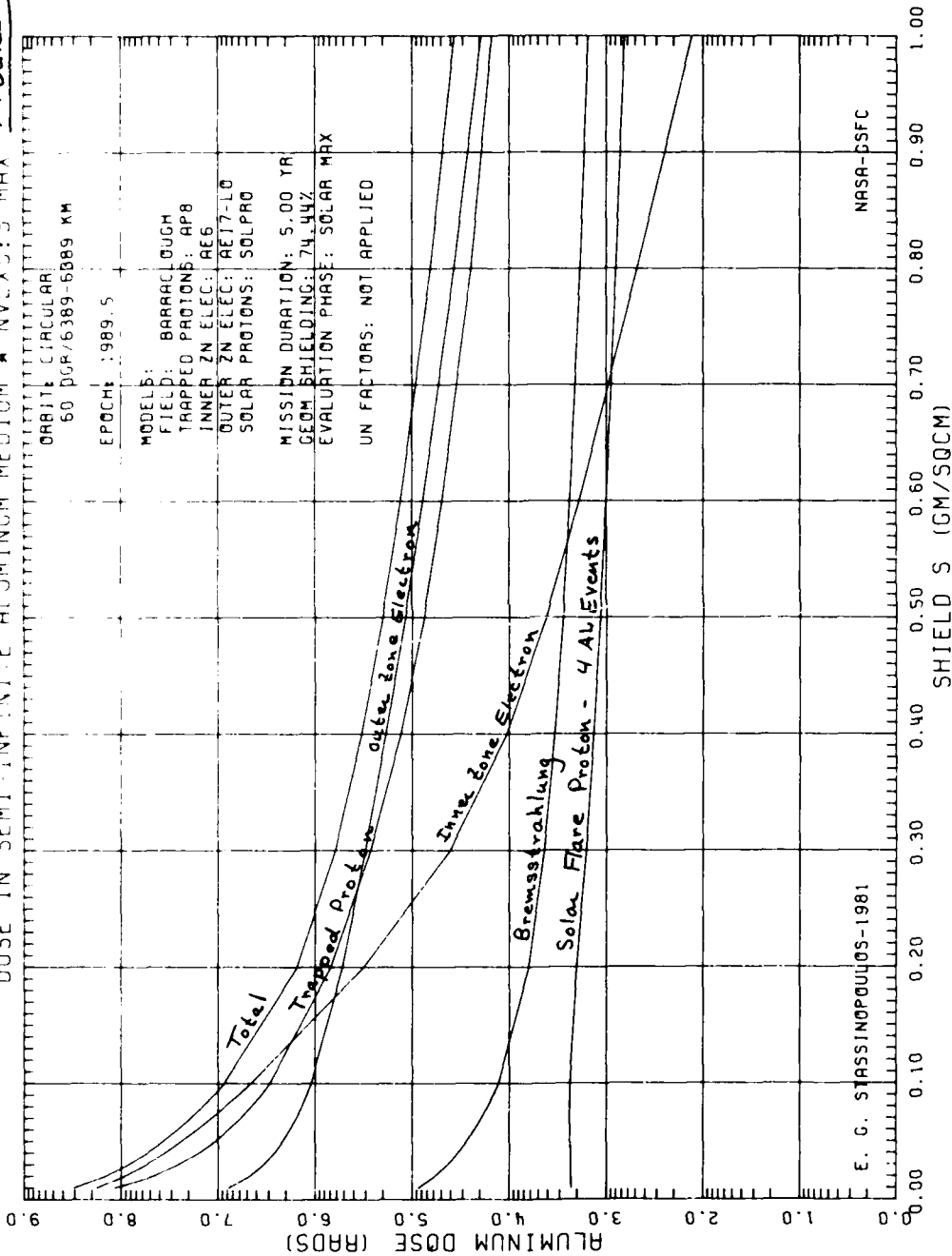
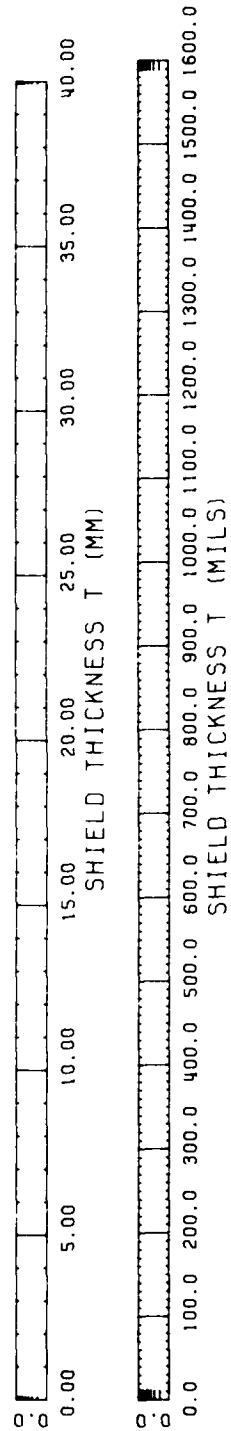
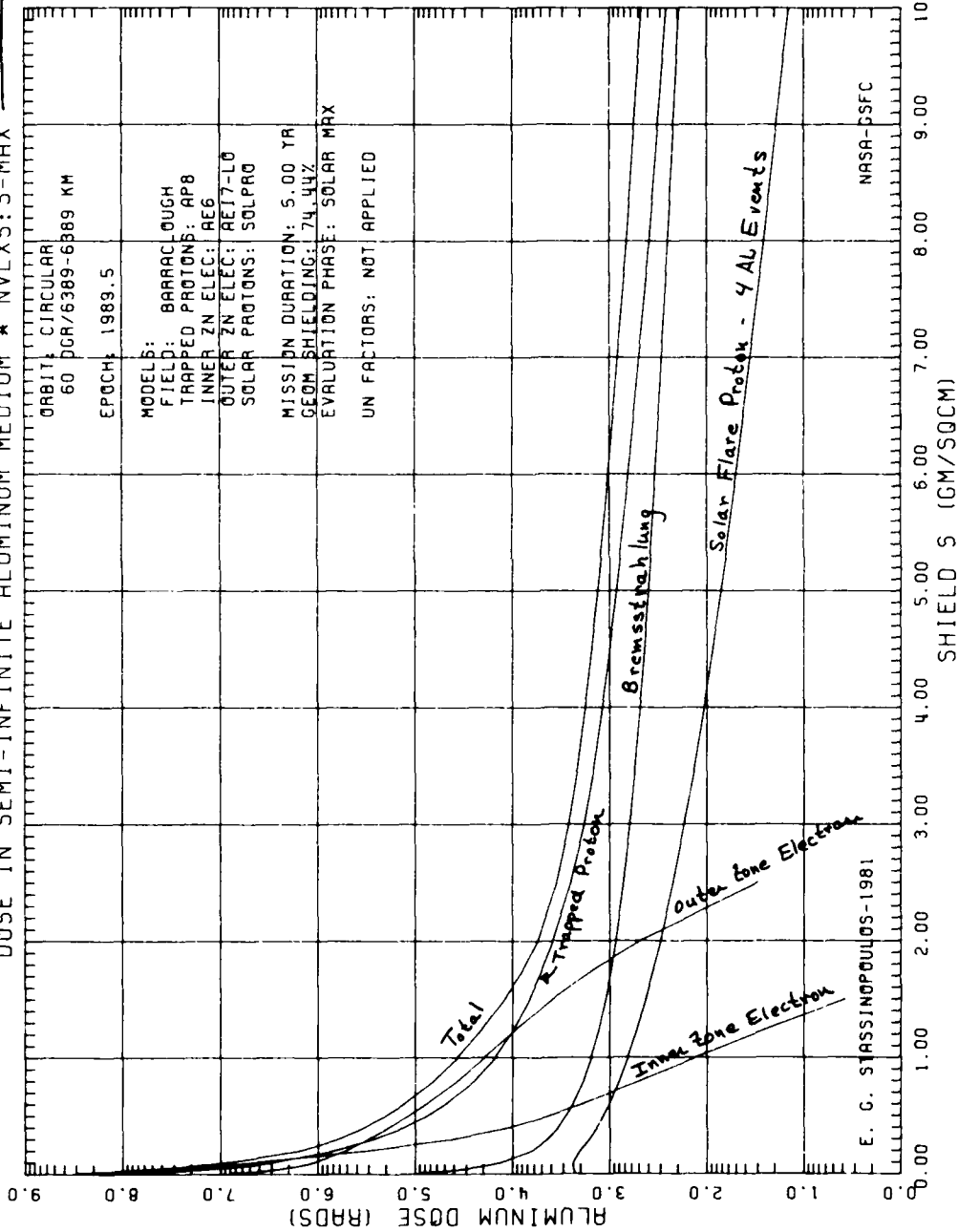


Figure 90

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX5: S-MAX



ORBIT: CIRCULAR  
60 DCR/6389-6889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARRACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AET7-L0  
SOLAR PROTONS: SOLPRO

MISSION DURATION: 5.00 YR  
GEOM SHALEDDING: 74.44%

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

E. G. STASSINOPOULOS-1981

Figure 91

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX5: S-MAX

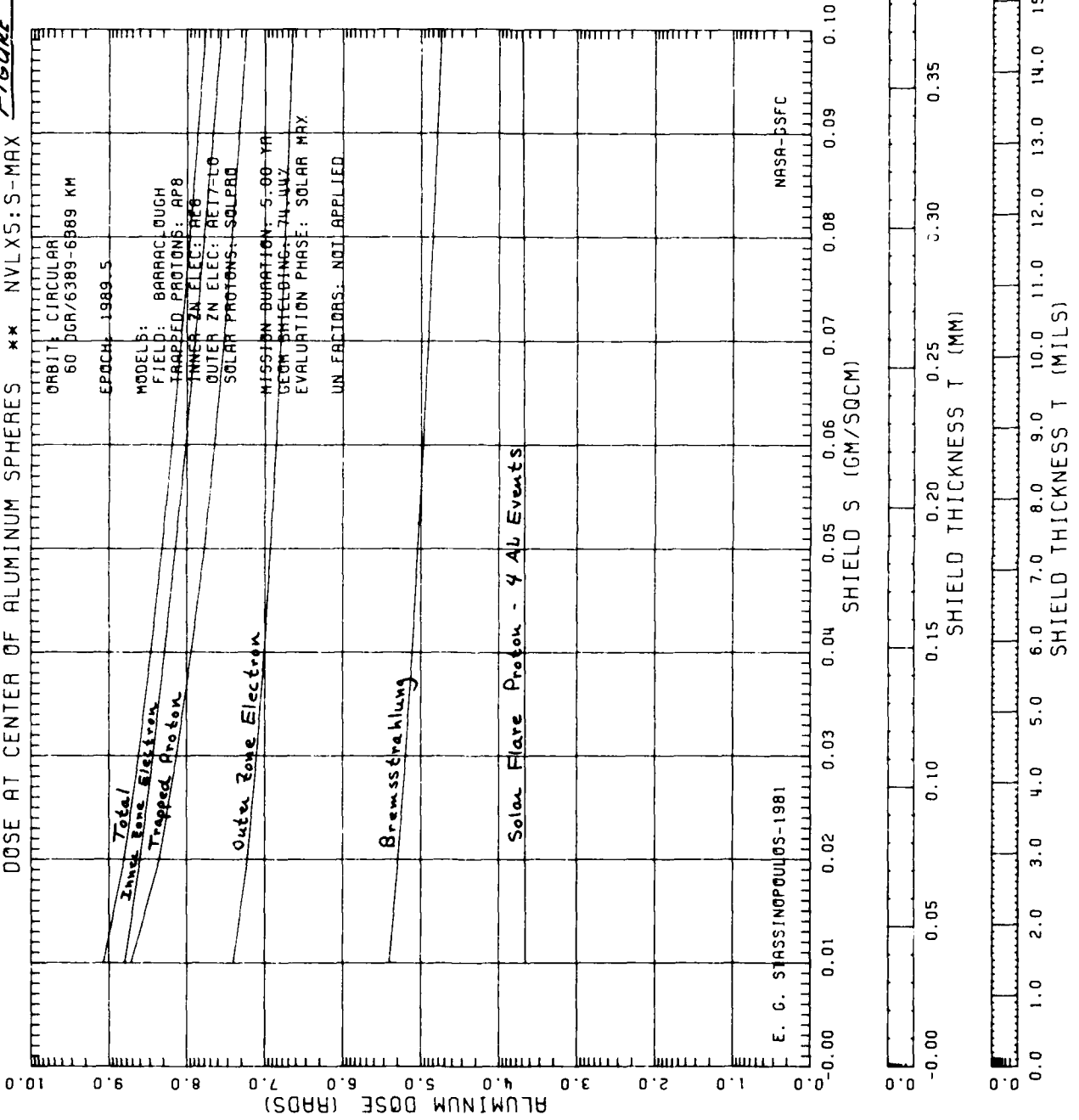


FIGURE 92

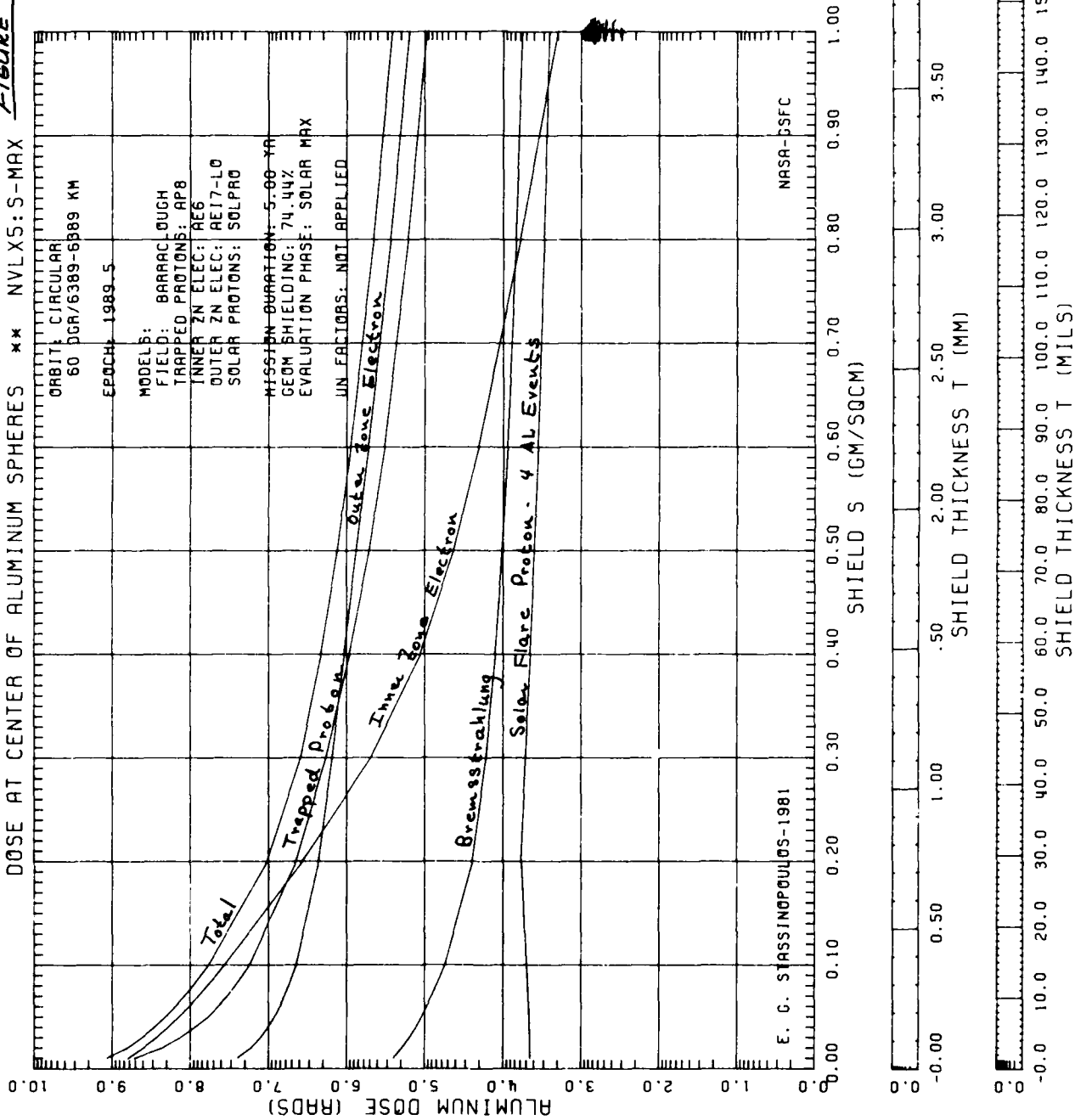
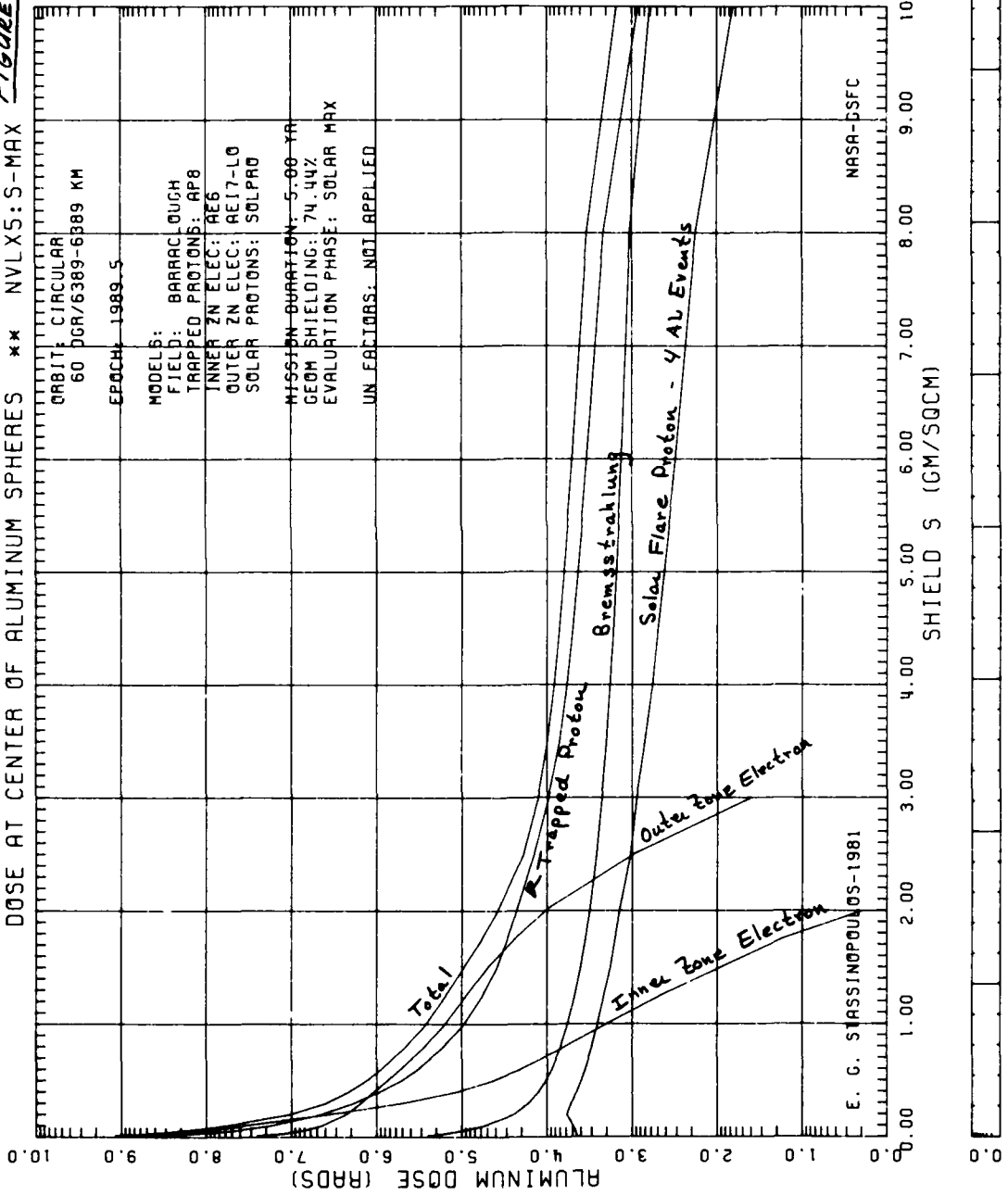


Figure 93

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX5: S-MAX



ORBIT: CIRCULAR  
60 DCR/6389-6889 KM  
EPOCH: 1989.5

MODELS:  
FIELD: BARRACLOUGH  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: RE6  
OUTER ZN ELEC: RE17-LO  
SOLAR PROTONS: SOLPRO

MISSION DURATION: 5.00 YR  
GEOM SHIELDING: 74.44%  
EVALUATION PHASE: SOLAR MAX  
UN FACTORS: NOT APPLIED

SHIELD THICKNESS T (MM)  
0.00 5.00 10.00 15.00 20.00 25.00 30.00 35.00 40.00

SHIELD THICKNESS T (MILS)  
0.0 100.0 200.0 300.0 400.0 500.0 600.0 700.0 800.0 900.0 1000.0 1100.0 1200.0 1300.0 1400.0 1500.0 1600.0

**FIGURE 94**  
**DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX6: S-MAX**

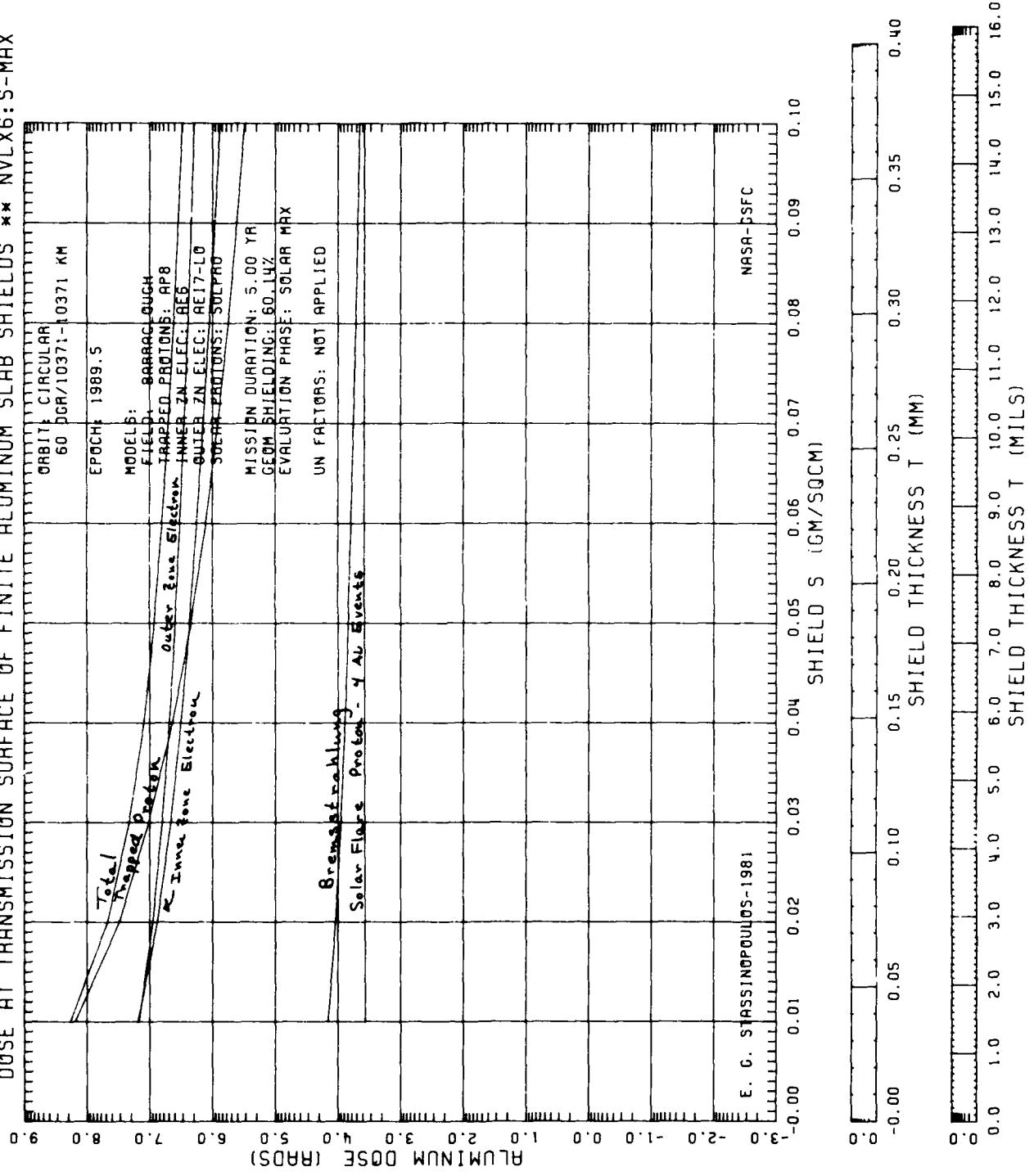




Figure 95

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX6: S-MAX

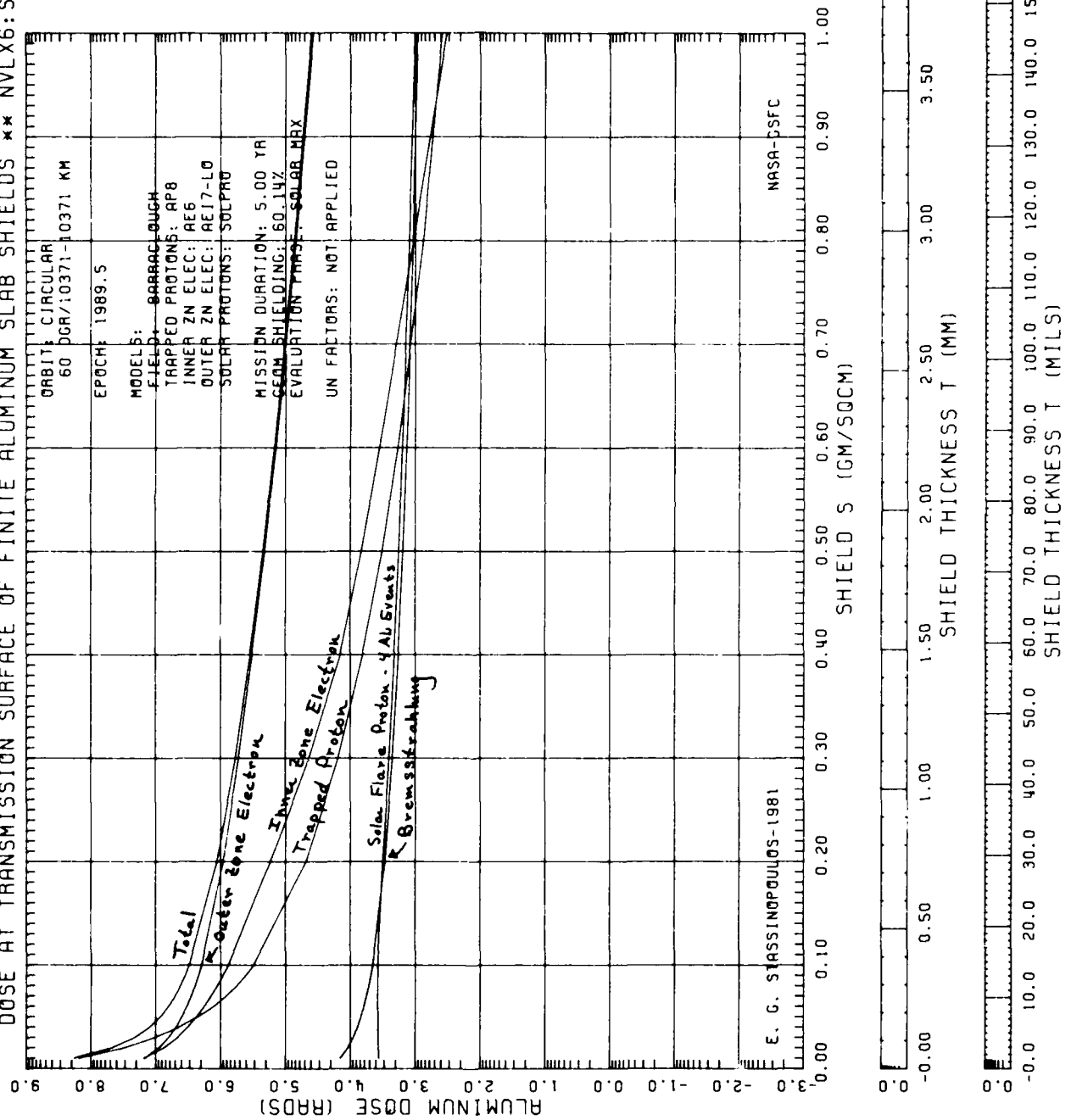


Figure 96

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS \*\* NVLX6: S-MAX

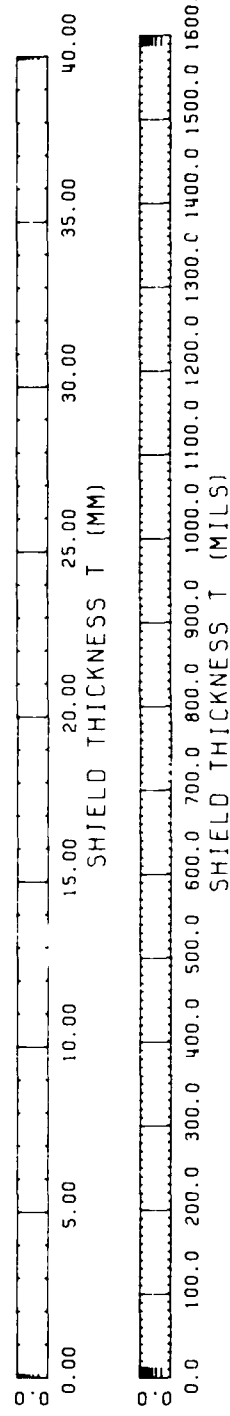
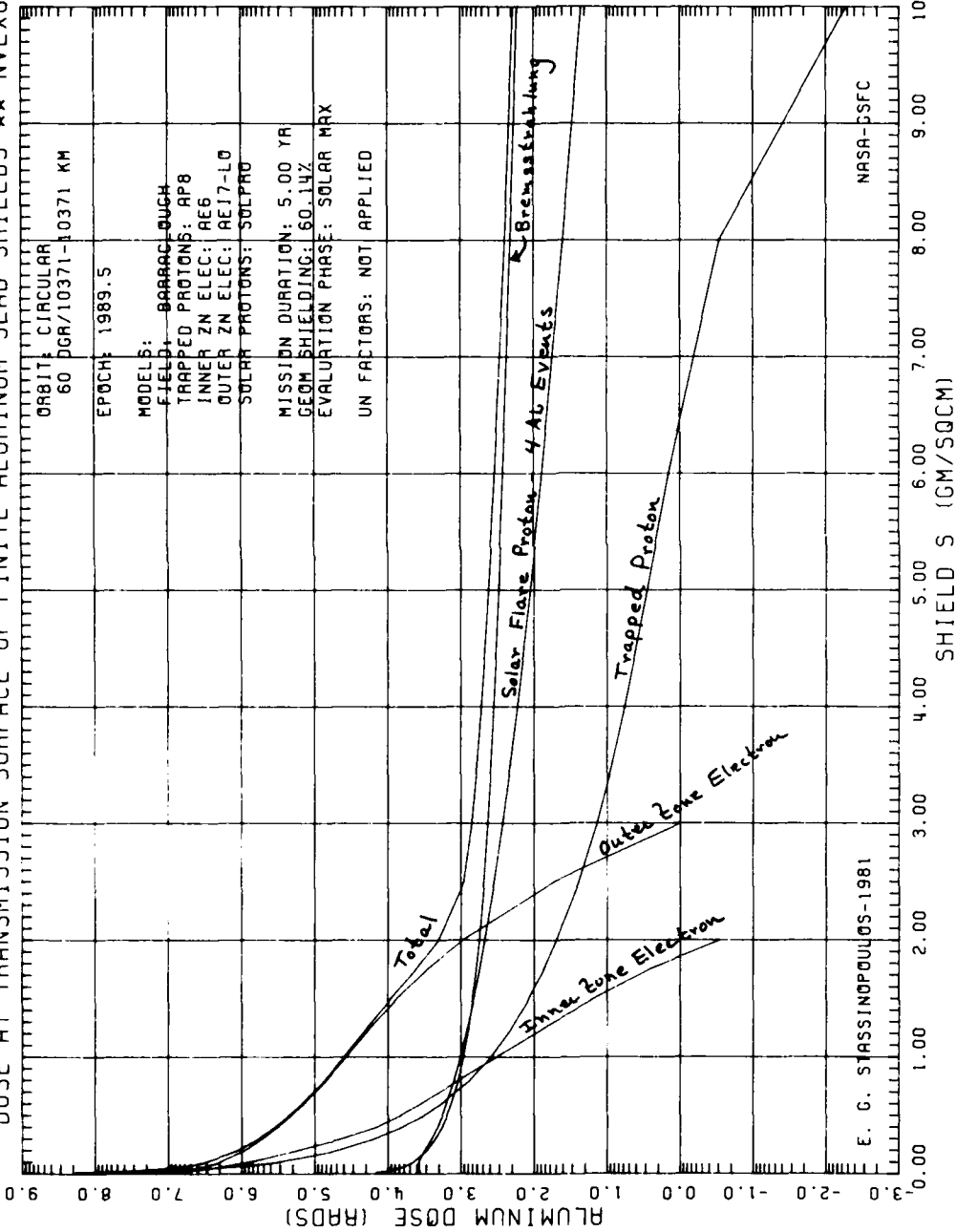


Figure 97

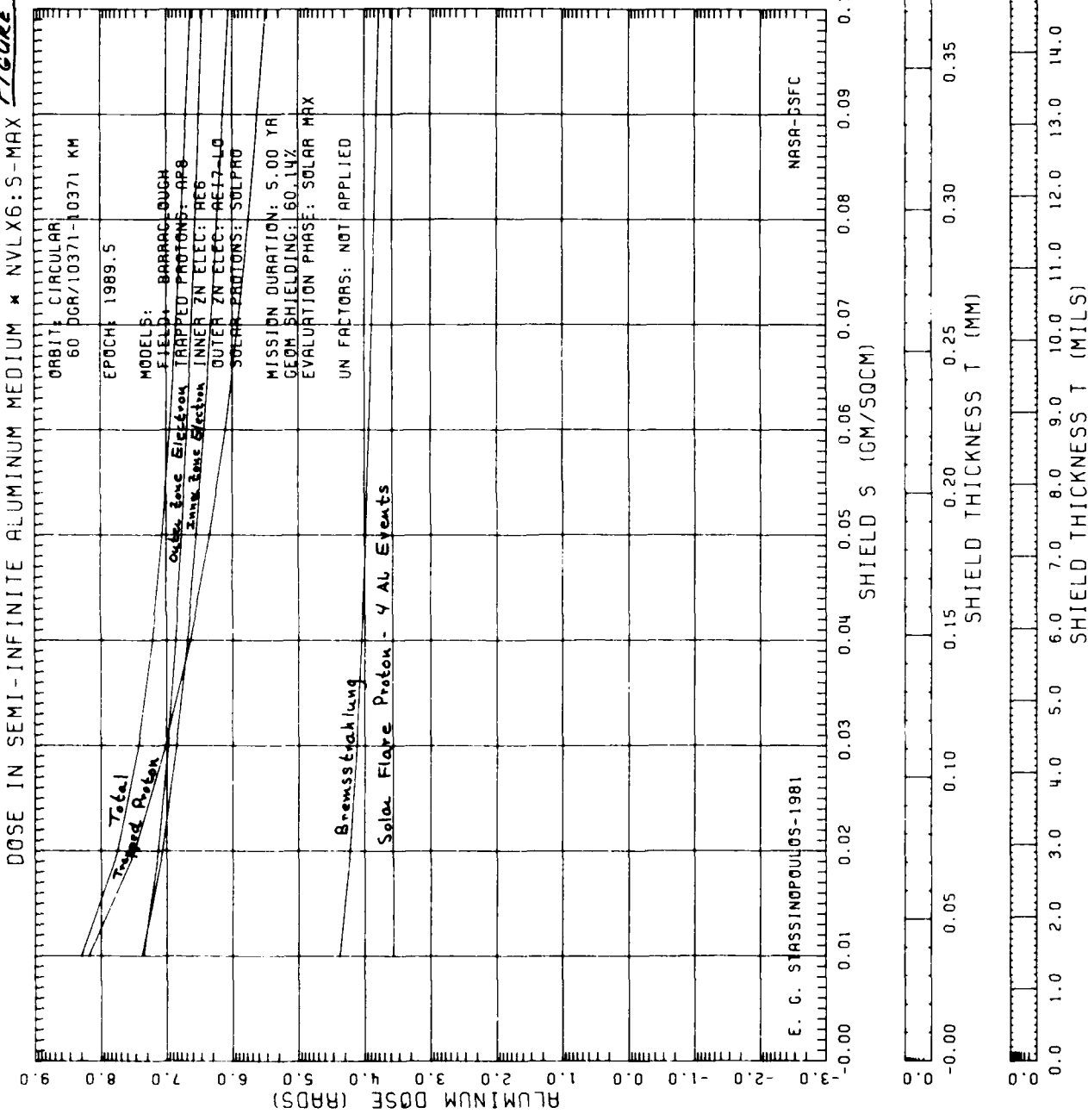
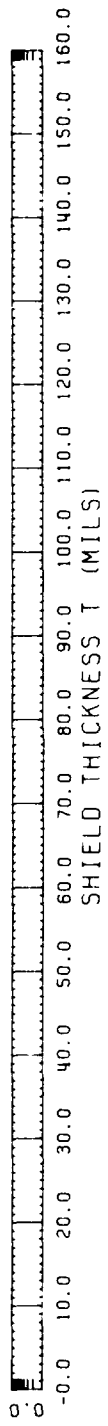
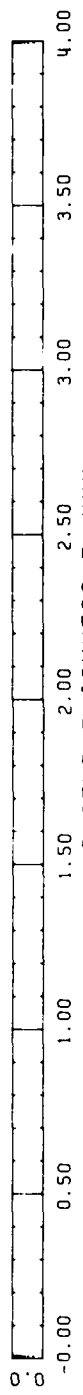
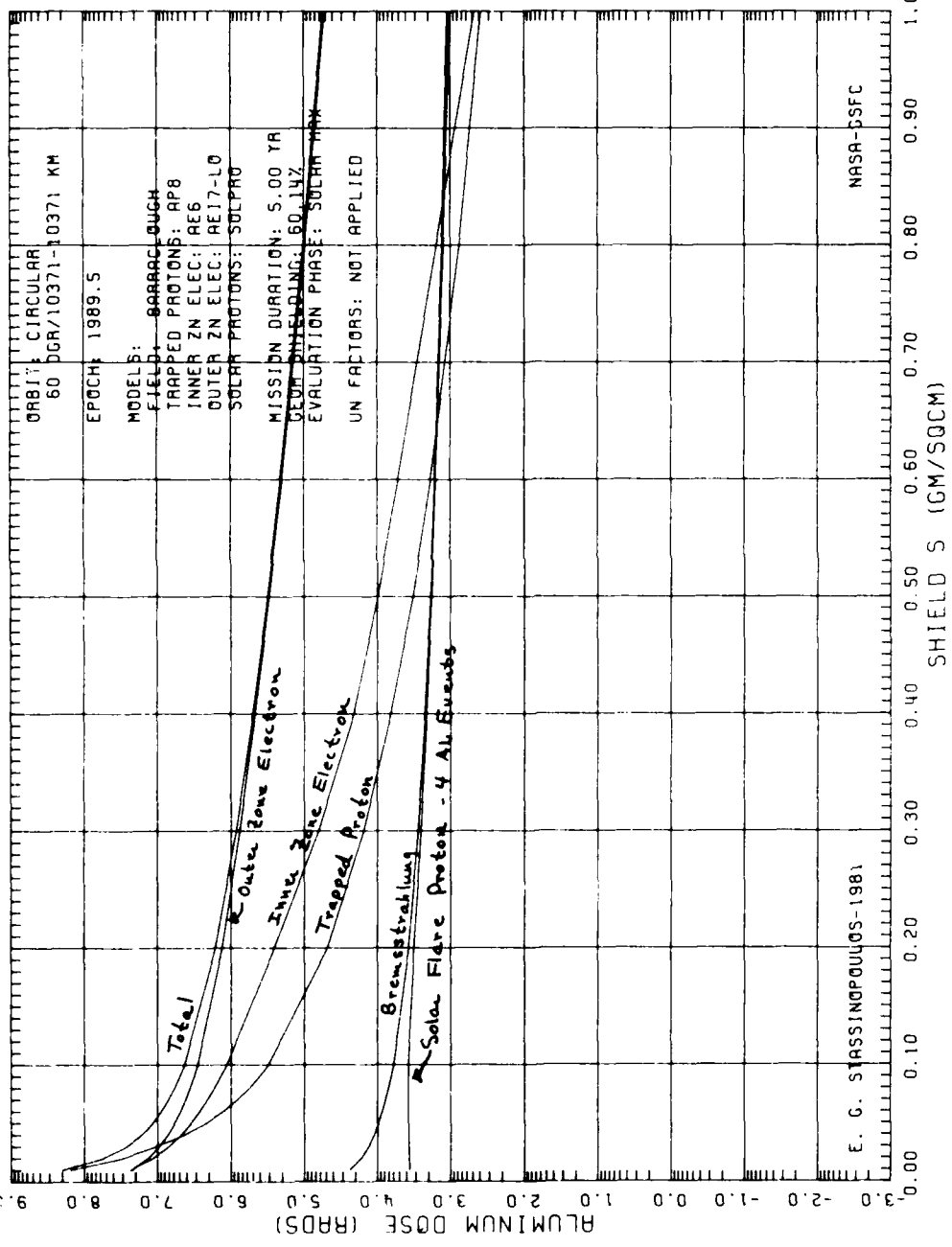
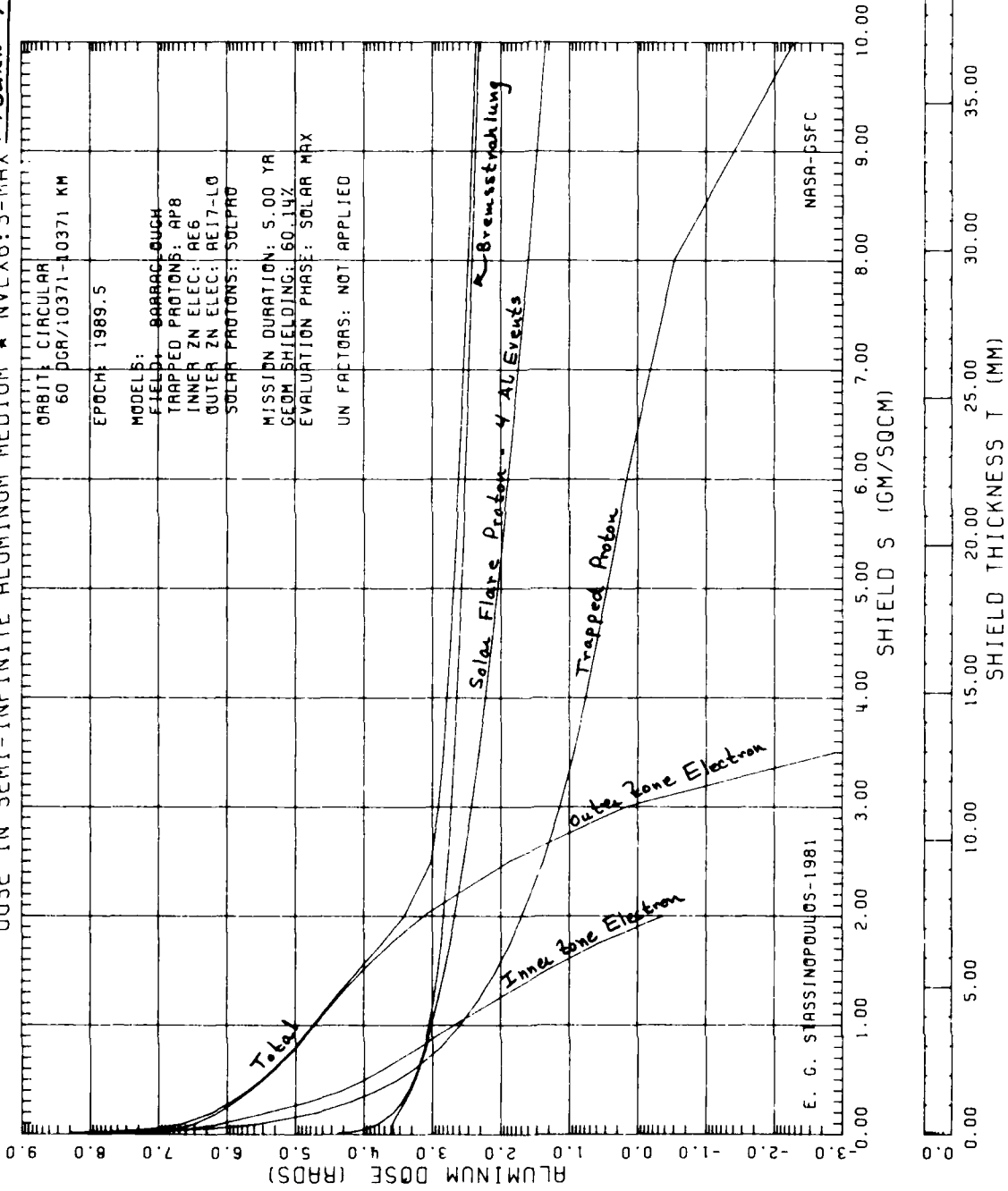


FIGURE 98

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX6:S-MAX



DOSE IN SEMI-INFINITE ALUMINUM MEDIUM \* NVLX6:S-MAX *Figure 99*



0.0 100.0 200.0 300.0 400.0 500.0 600.0 700.0 800.0 900.0 1000.0 1100.0 1200.0 1300.0 1400.0 1500.0 1600.0

SHIELD THICKNESS T (MILS)

Figure 100

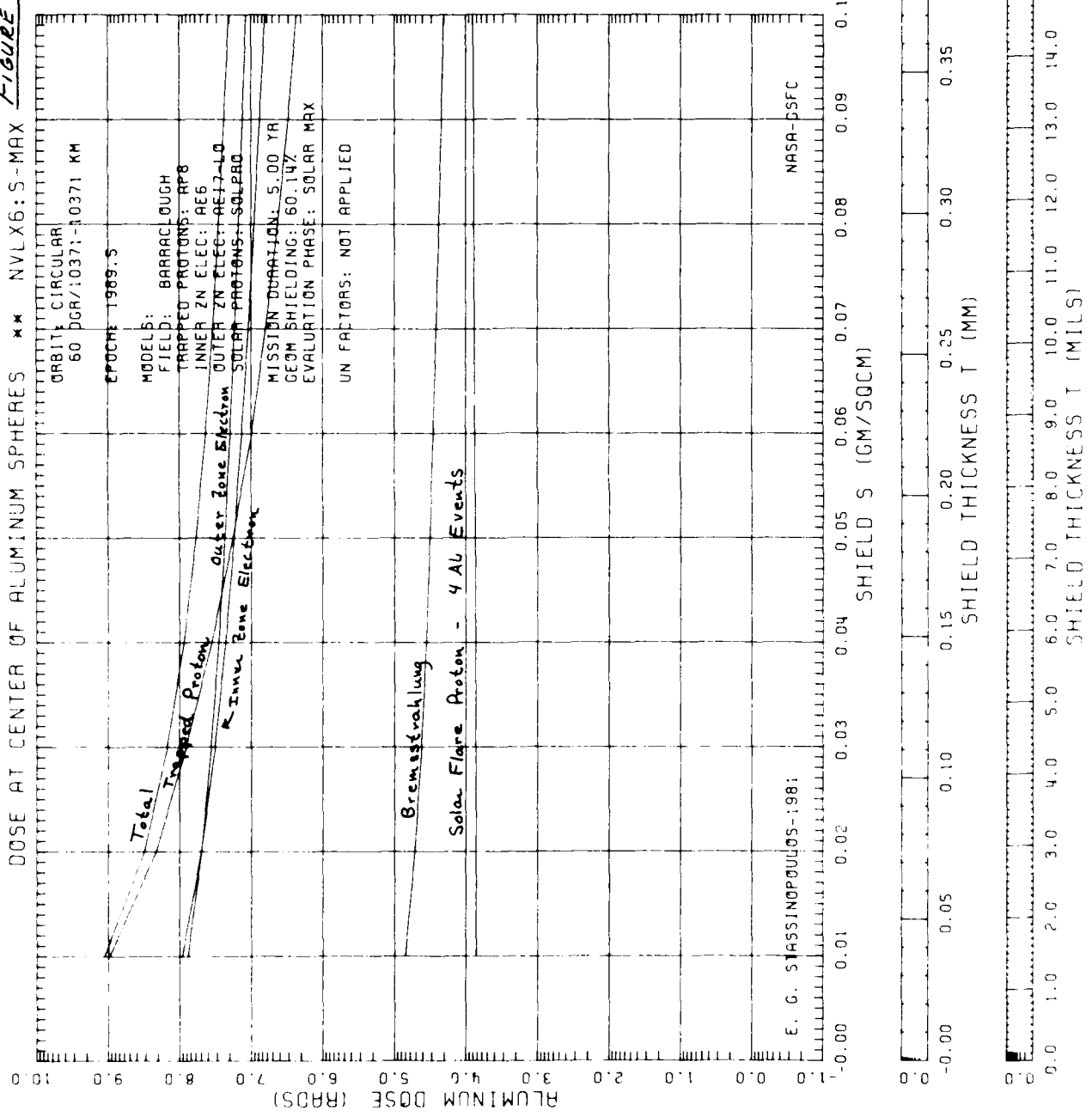
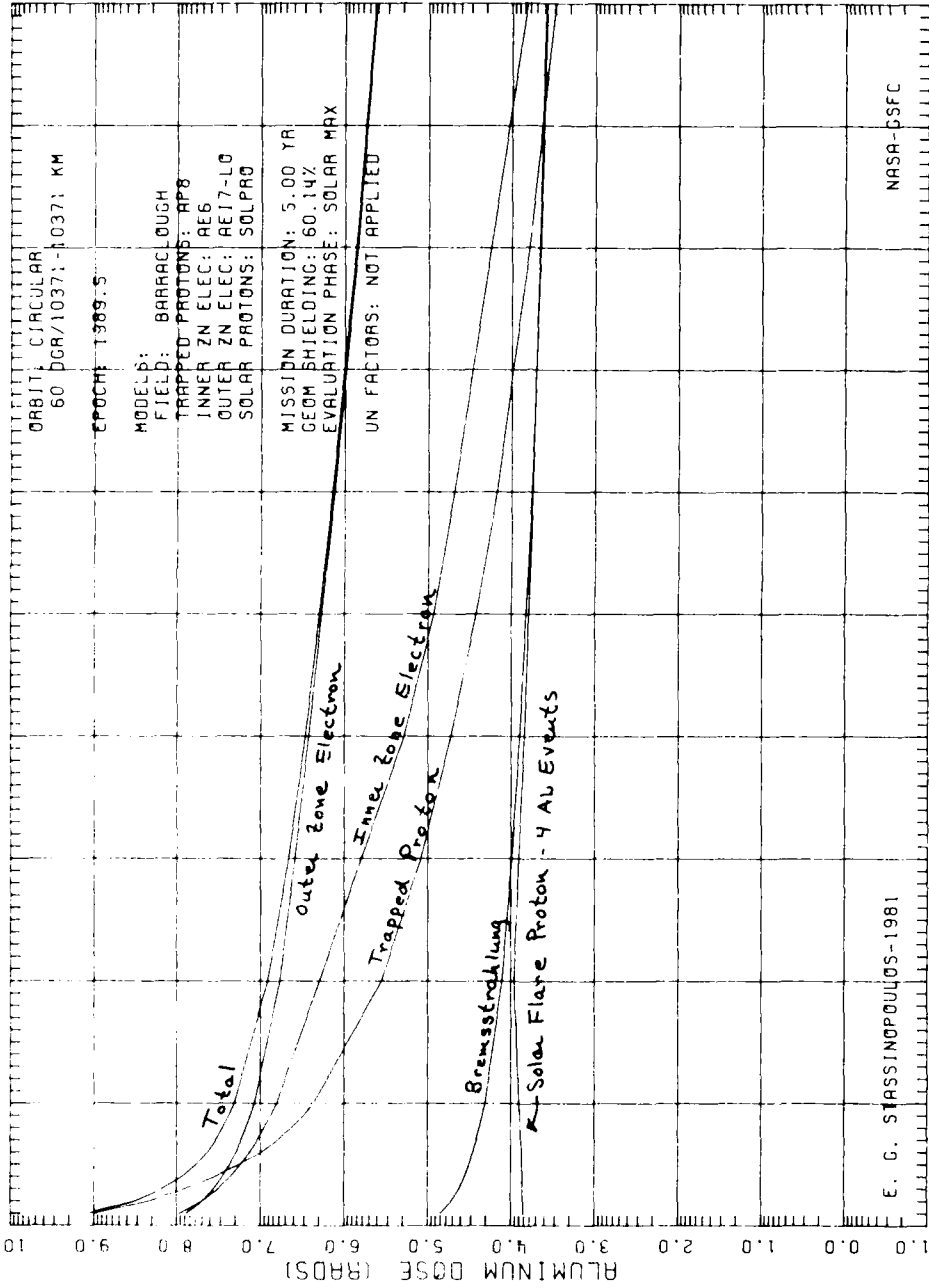


FIGURE 101

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX6: S-MAX



SHIELD S (GM/SQCM)

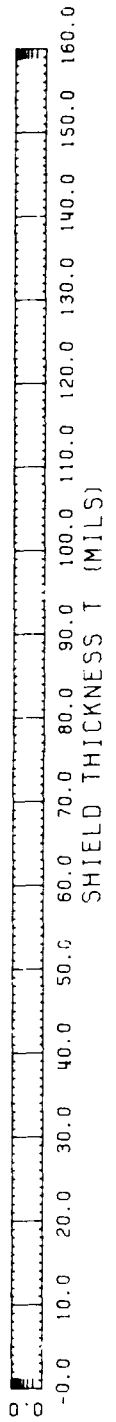
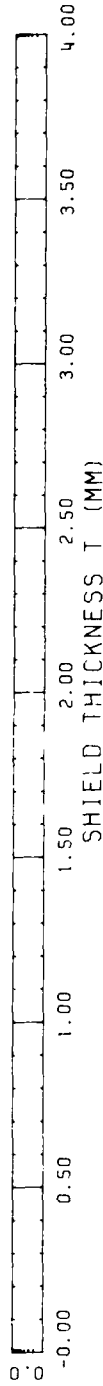
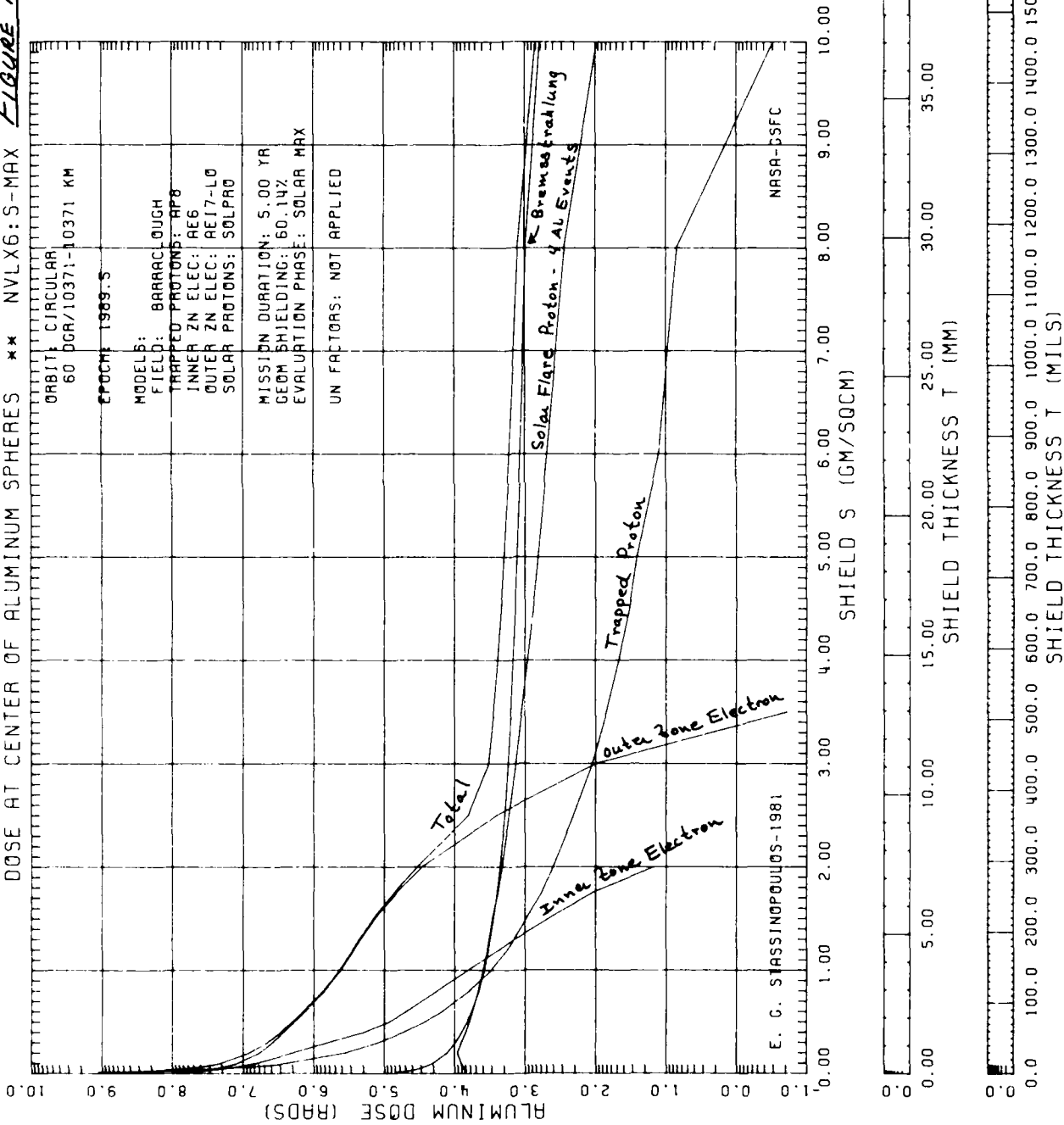
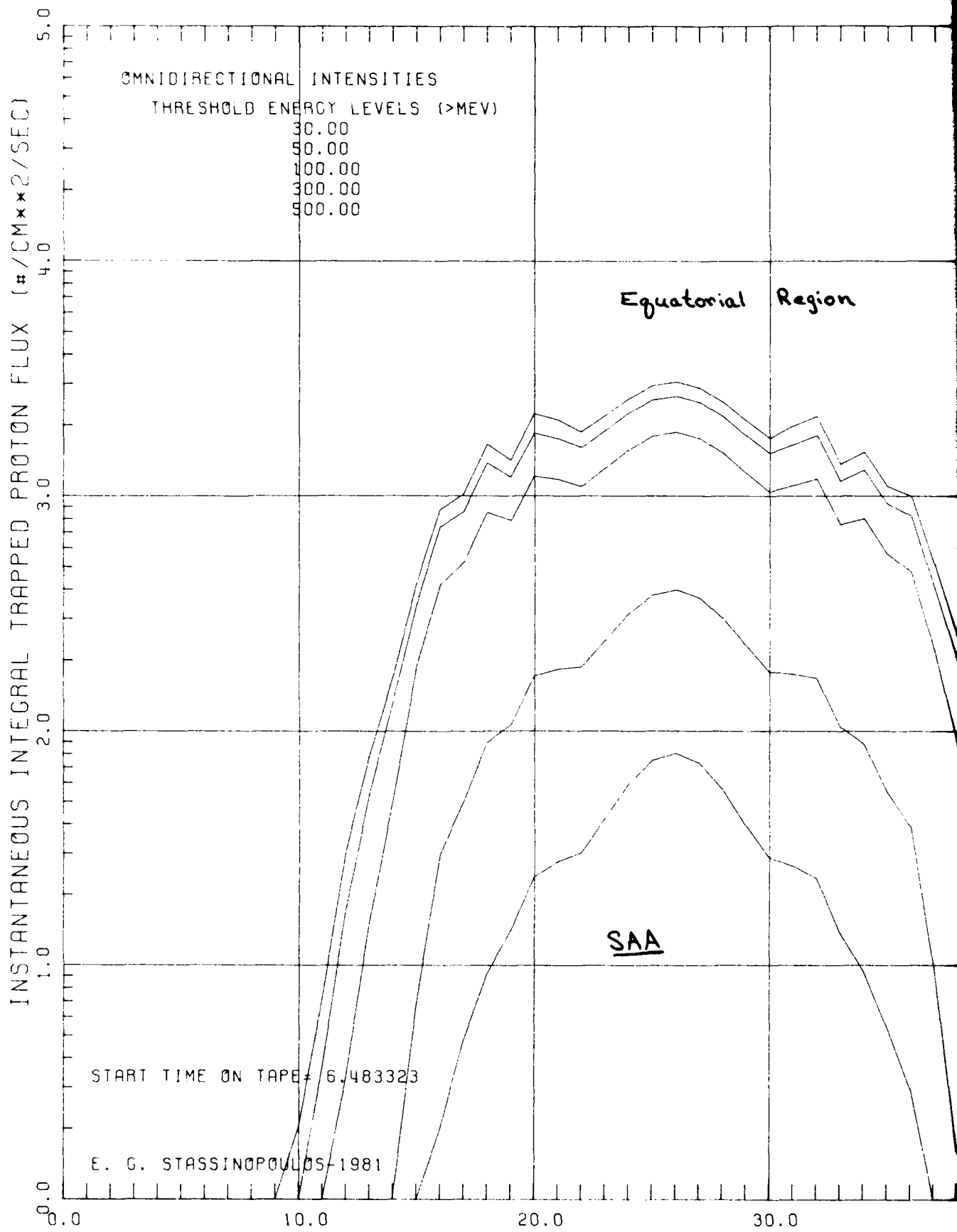


FIGURE 102

DOSE AT CENTER OF ALUMINUM SPHERES \*\* NVLX6:S-MAX







Region

30.0

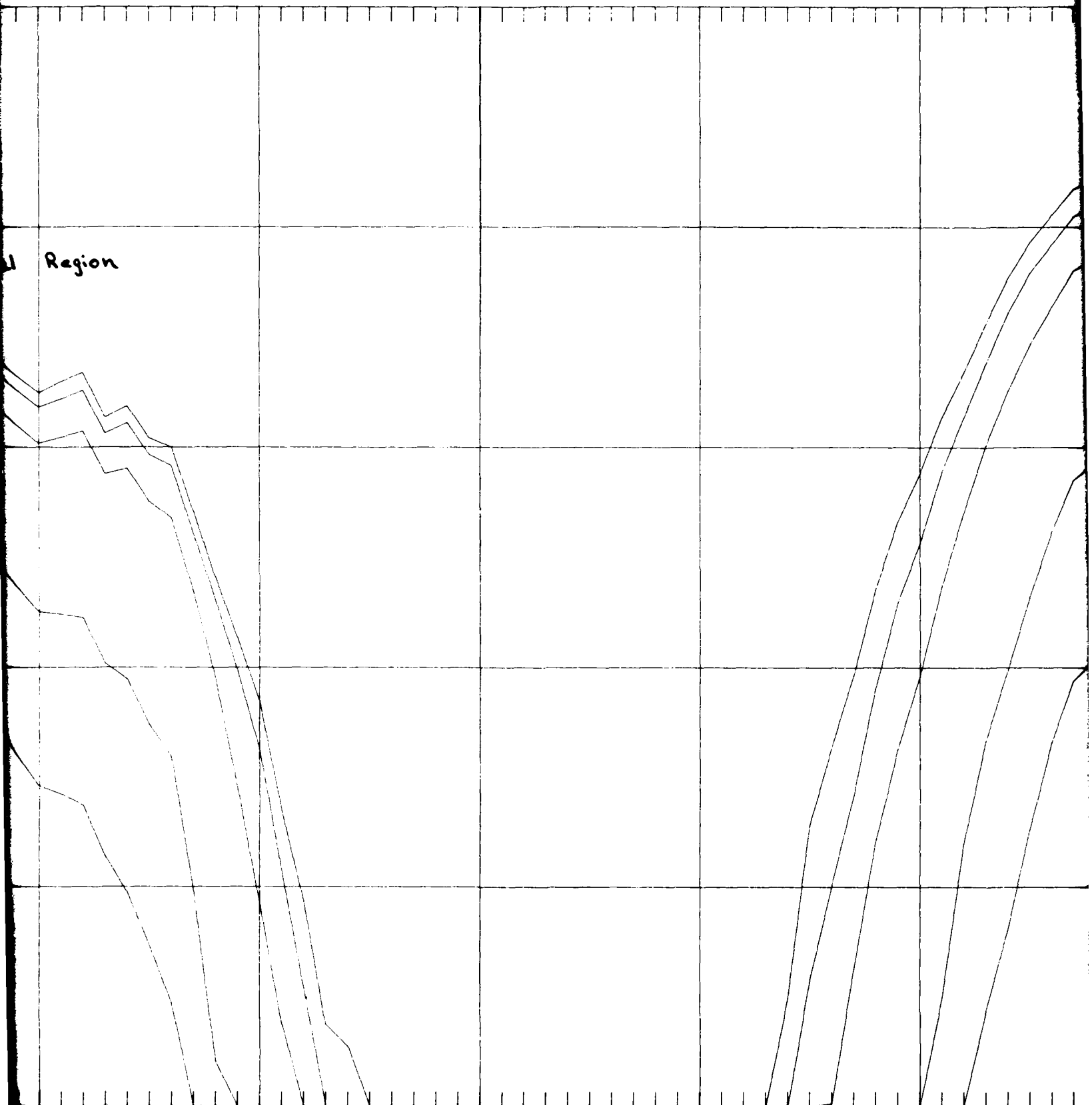
40.0

50.0

60.0

70.0

RELATIVE ORBIT TIME (MINUTES)



Equatorial Region

ORBIT: NAVELEX 1  
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L  
MISSION DURATION: 60.00  
EVALUATION PHASE: SOLAR

UN FACTORS: NOT APPLIED

SAA

STOP TIME ON TAPE = 8.441

NASA

MINUTES) 70.0 80.0 90.0 100.0 110.0

Figure 103

ORBIT: NAVELEX 1  
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

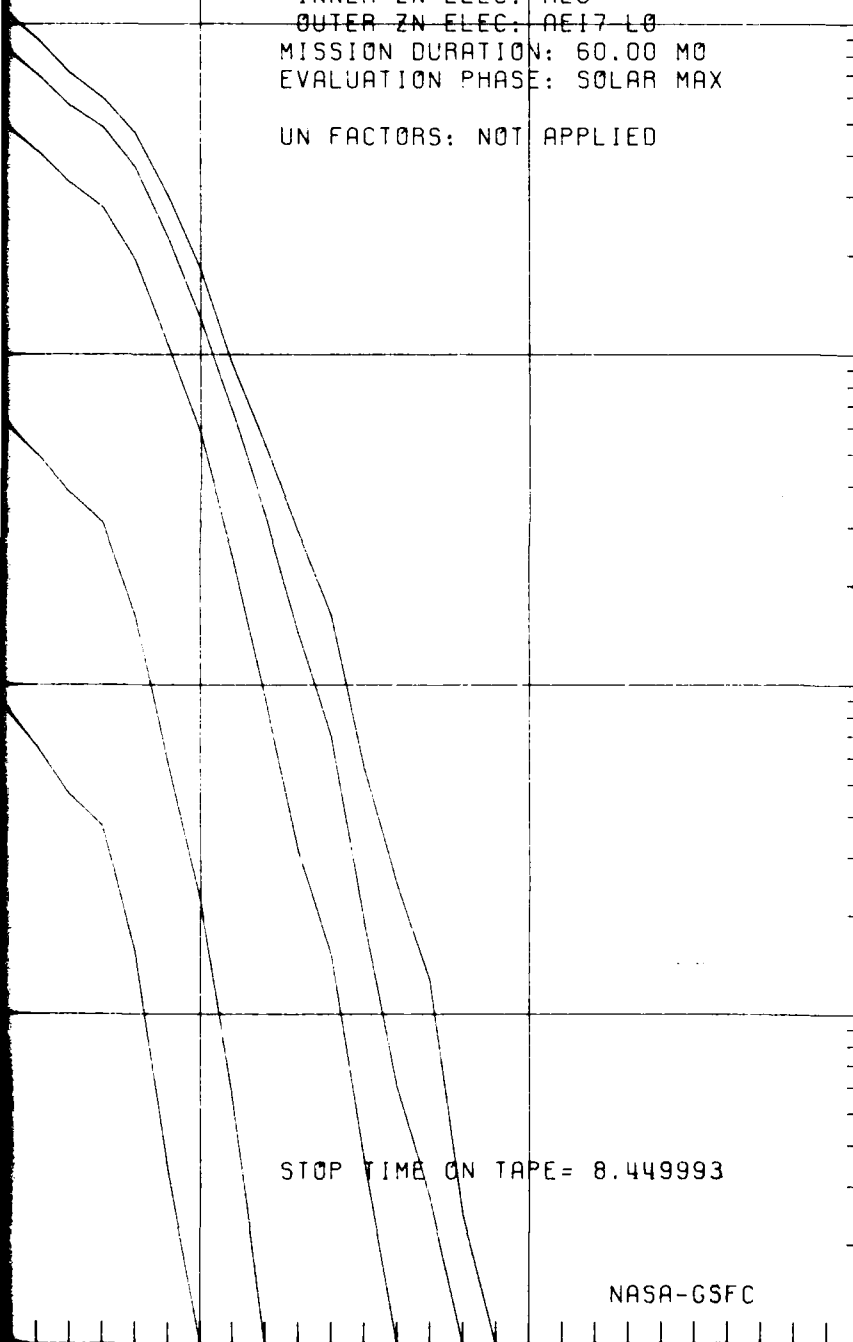
INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

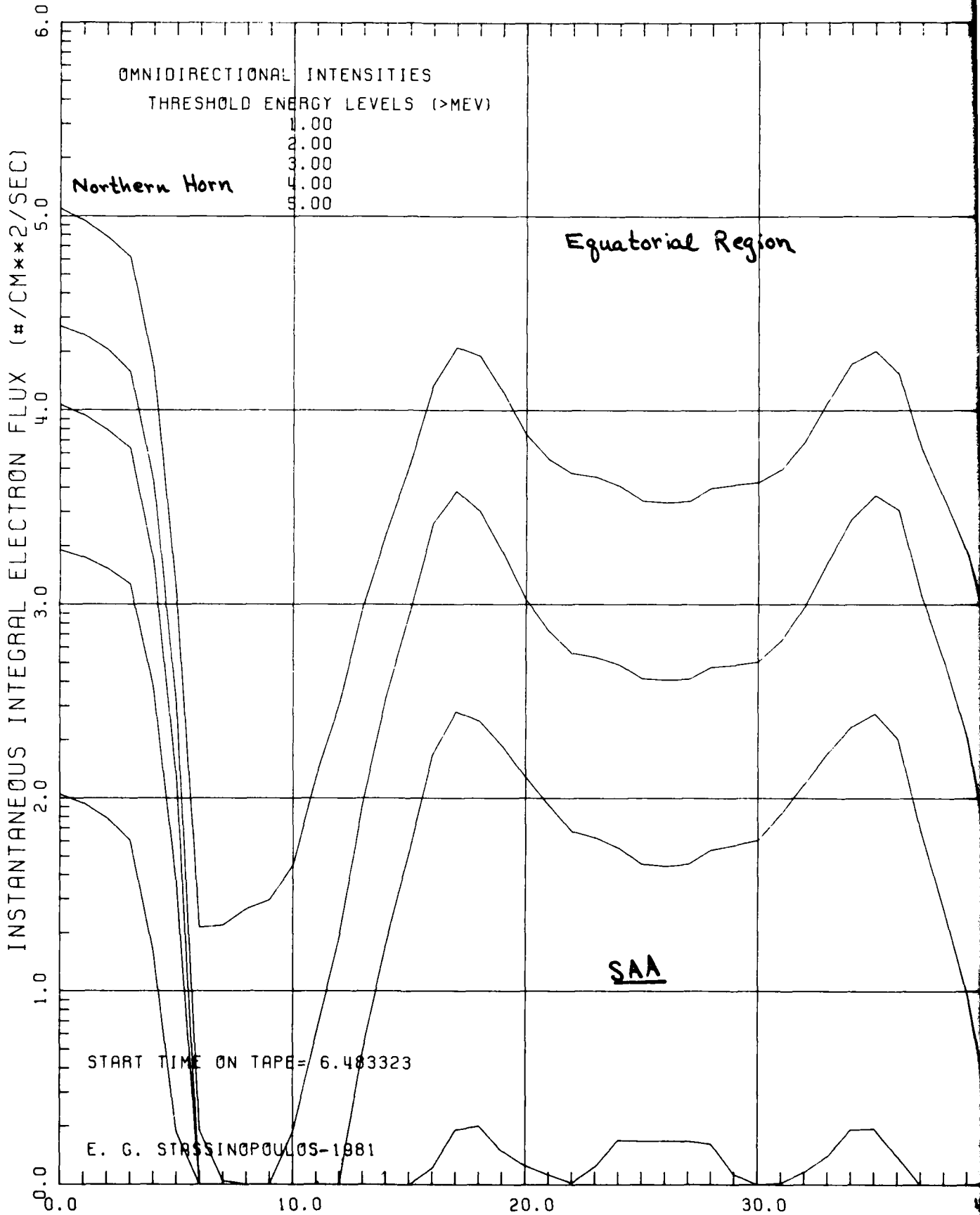
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

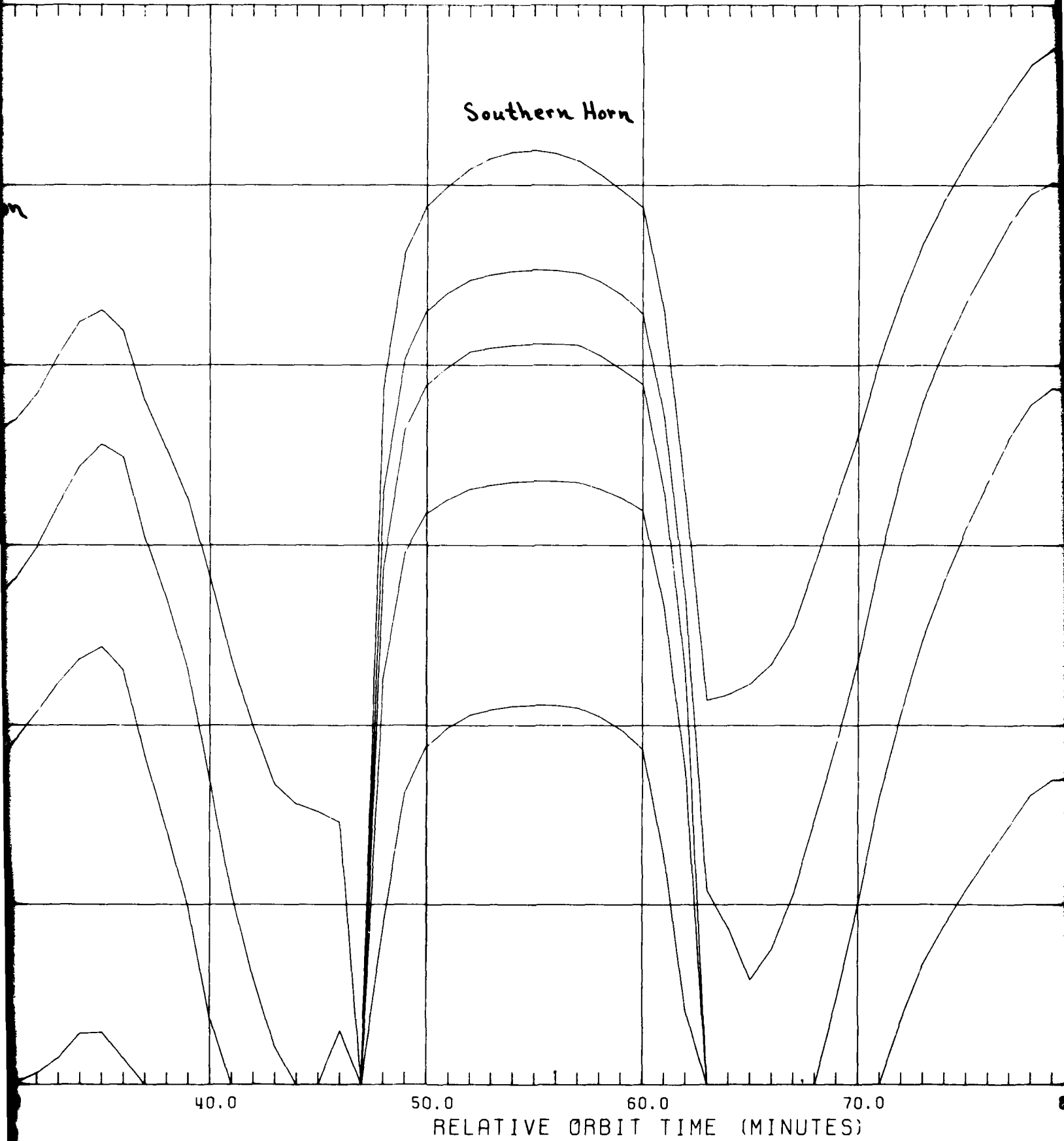


STOP TIME ON TAPE= 8.449993

NASA-GSFC



Southern Horn



3

# Equatorial Region

Figure

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

North

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-LO  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

SAA

STOP TIME ON TAPE = 8.449993

NASA-GS

70.0 80.0 90.0 100.0 110.0

UTES)

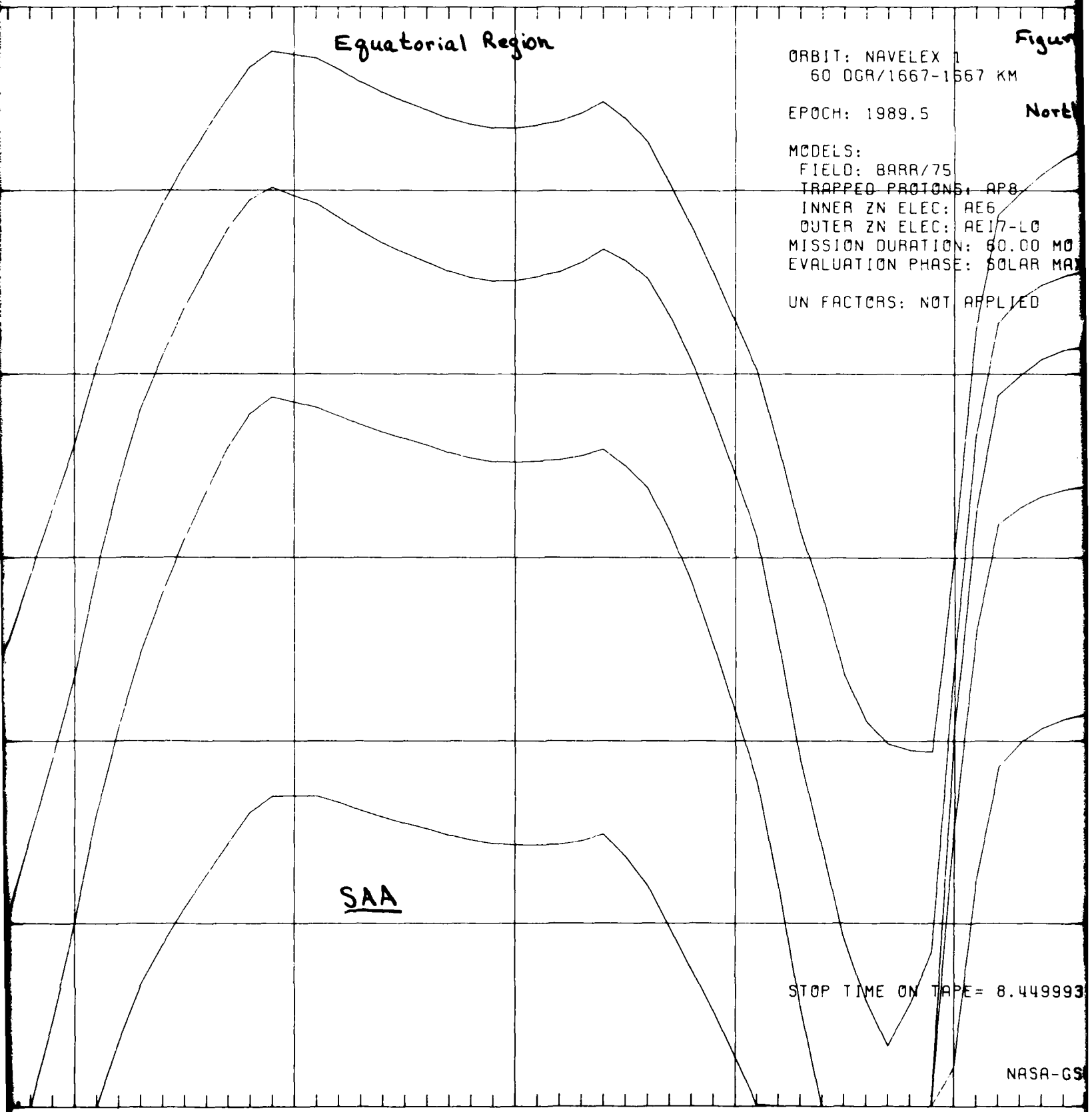


Figure 104

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

Northern Horn

MCDELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

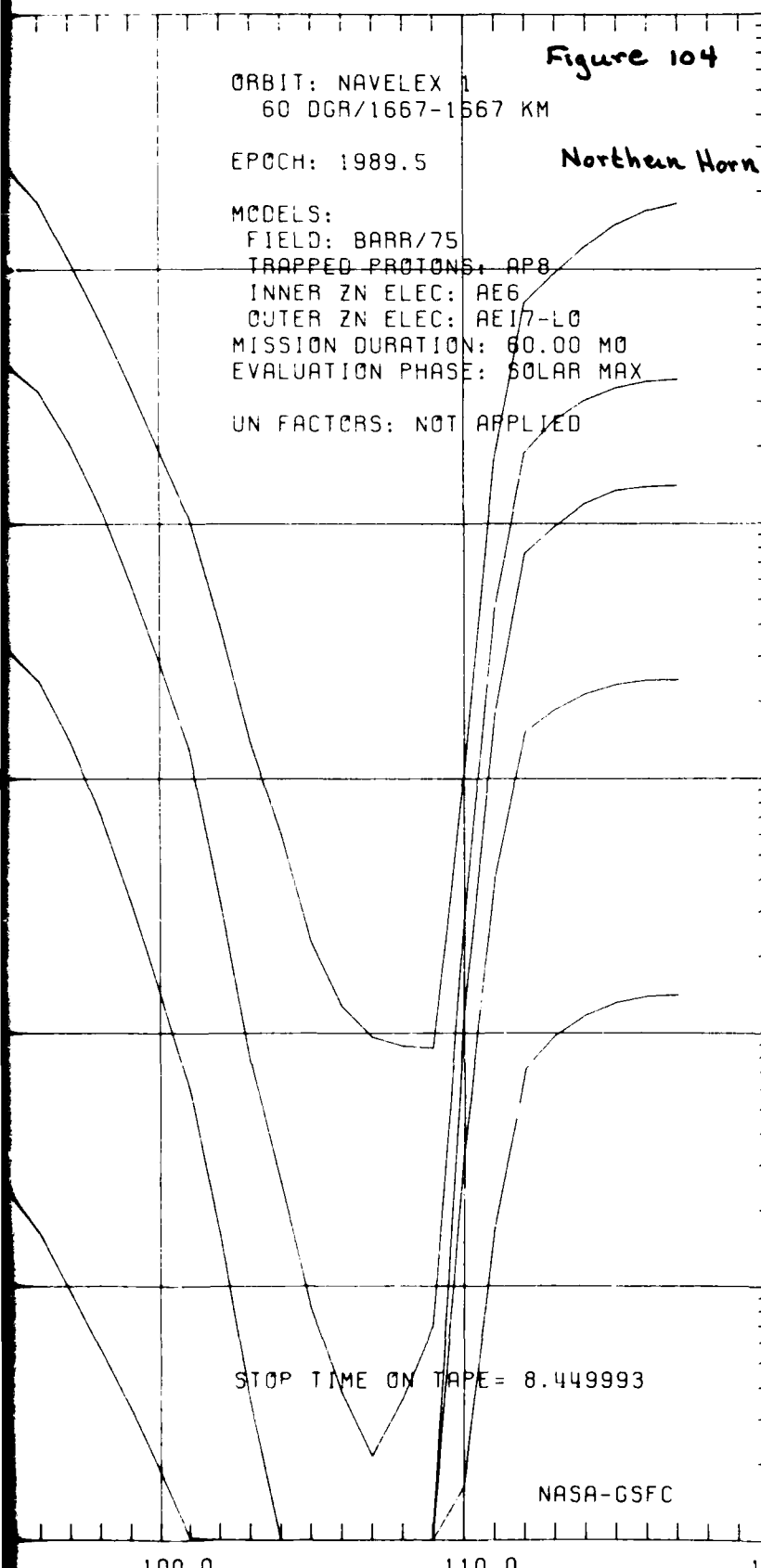
INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED



STOP TIME ON TAPE = 8.449993

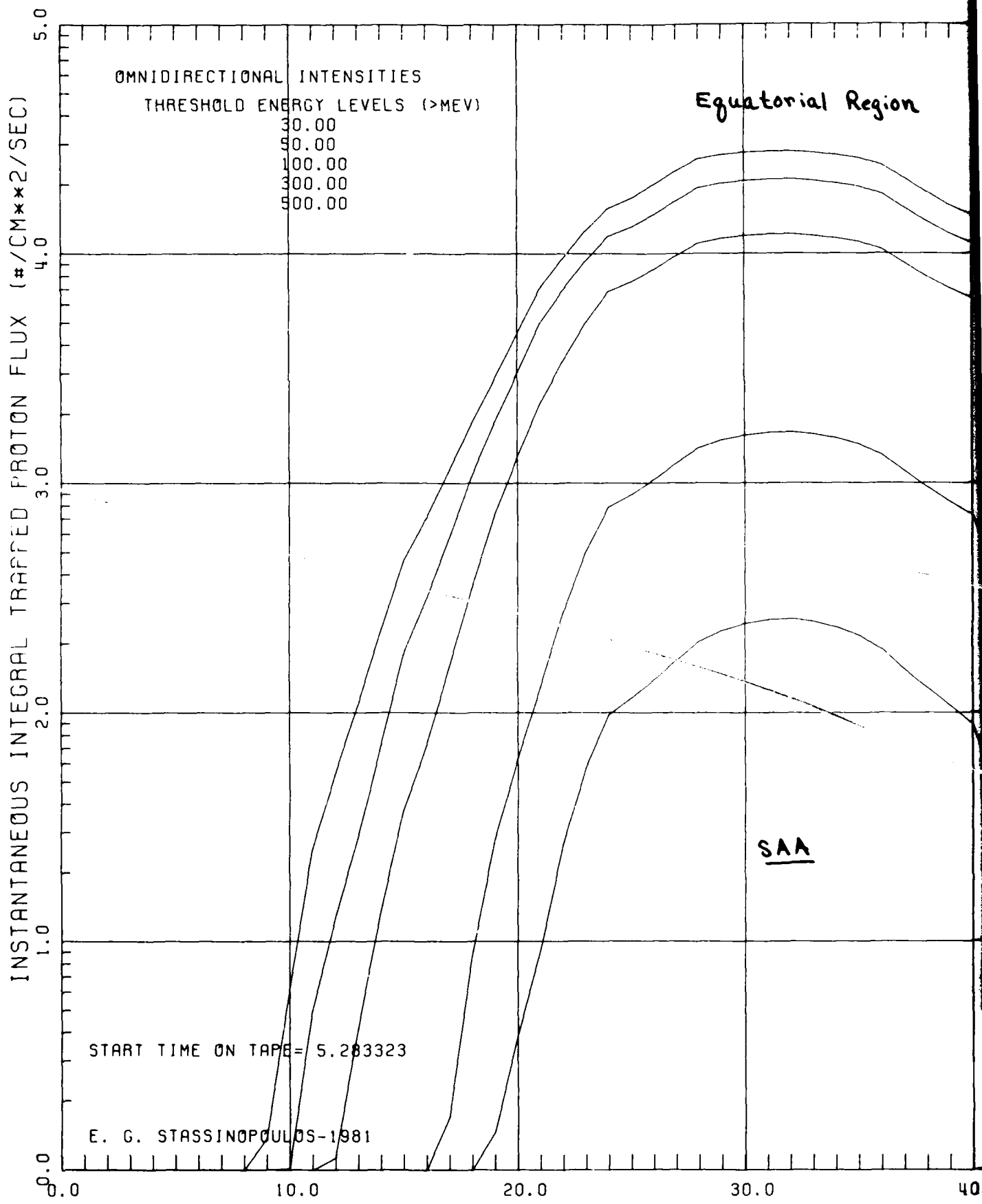
NASA-GSFC

100.0

110.0

120.0

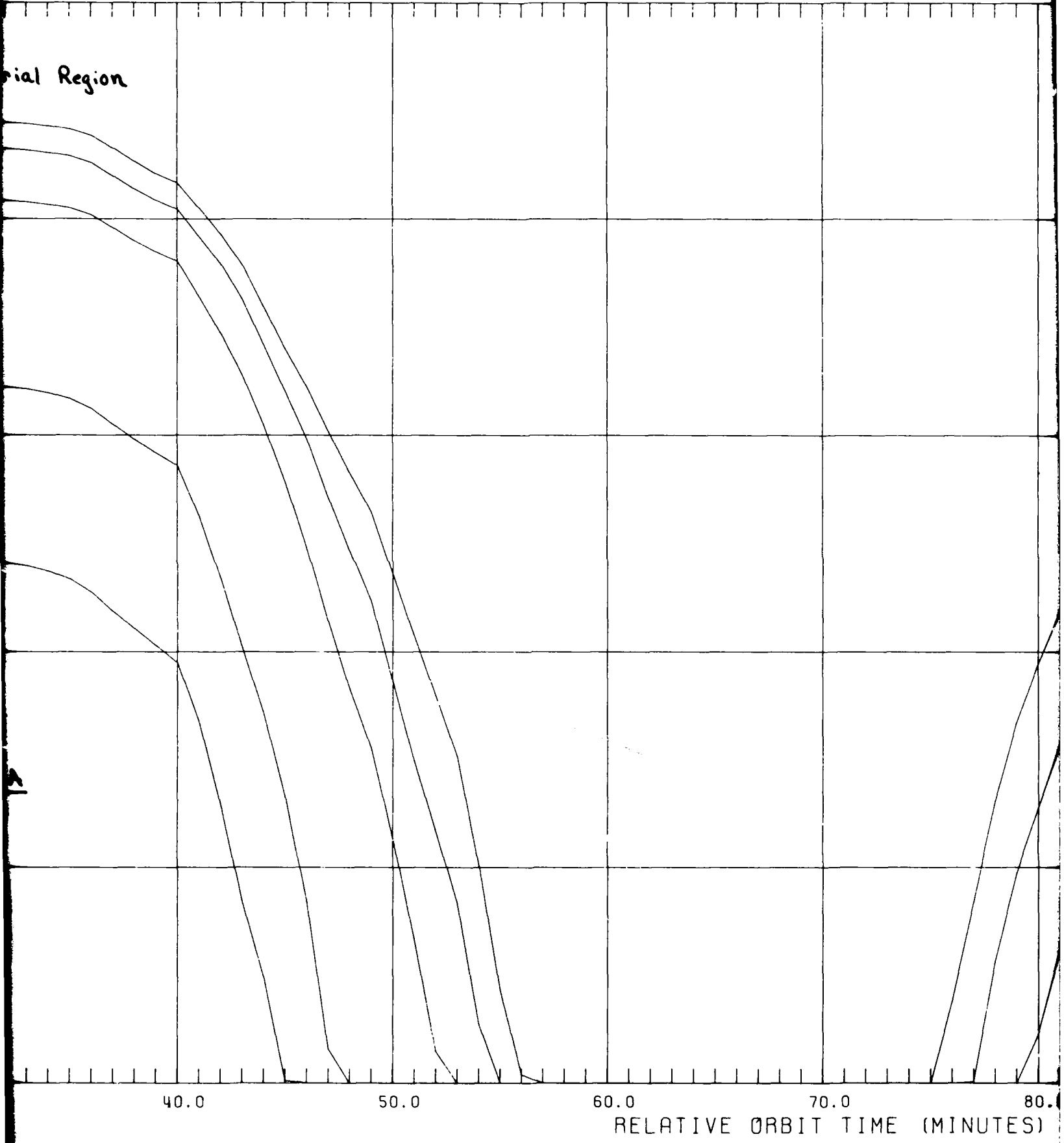




2

1

ial Region



RELATIVE ORBIT TIME (MINUTES)

Equatorial Region

SAA

80.0 90.0 100.0 110.0 120.0  
ME (MINUTES)

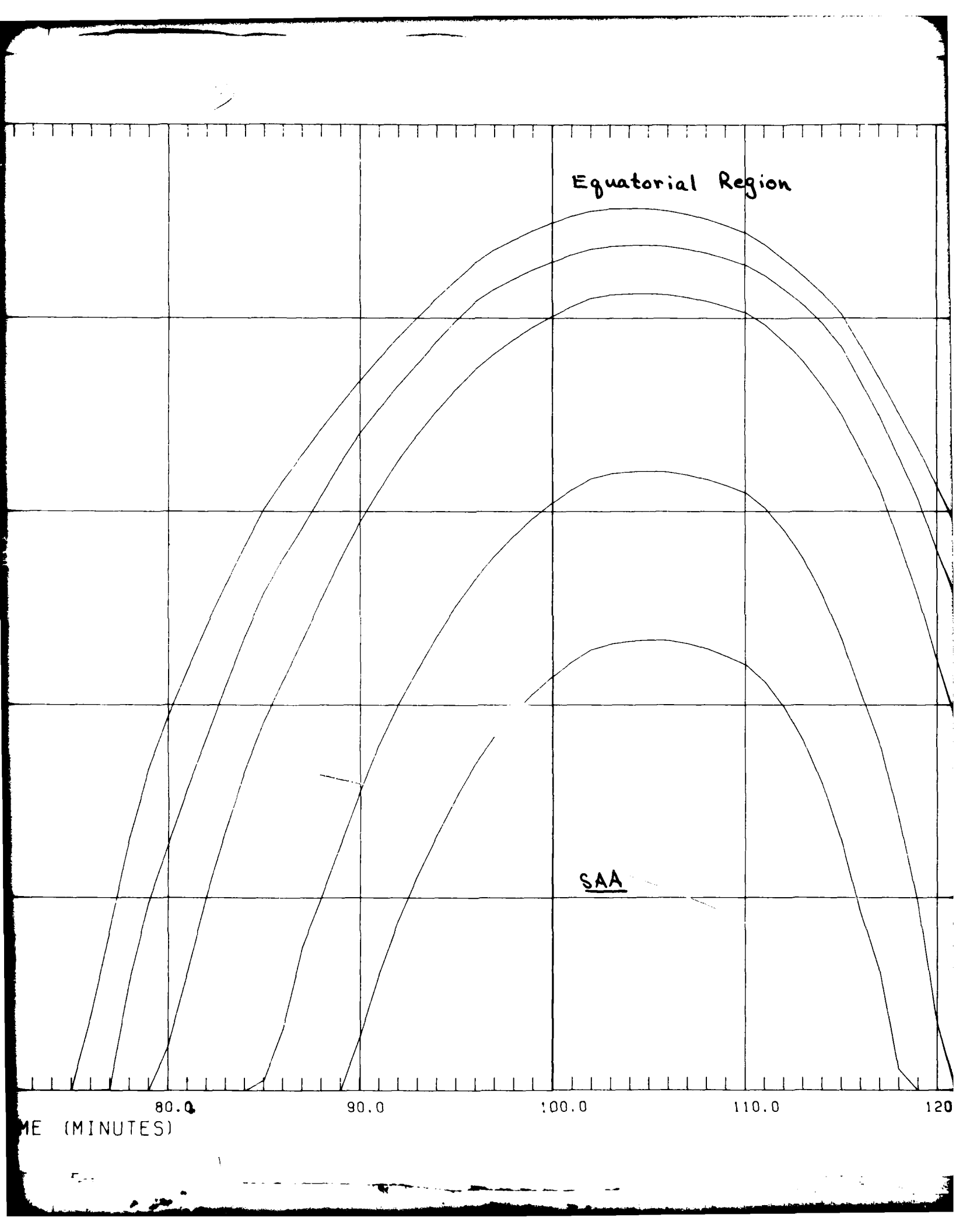


Figure 105

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AEG

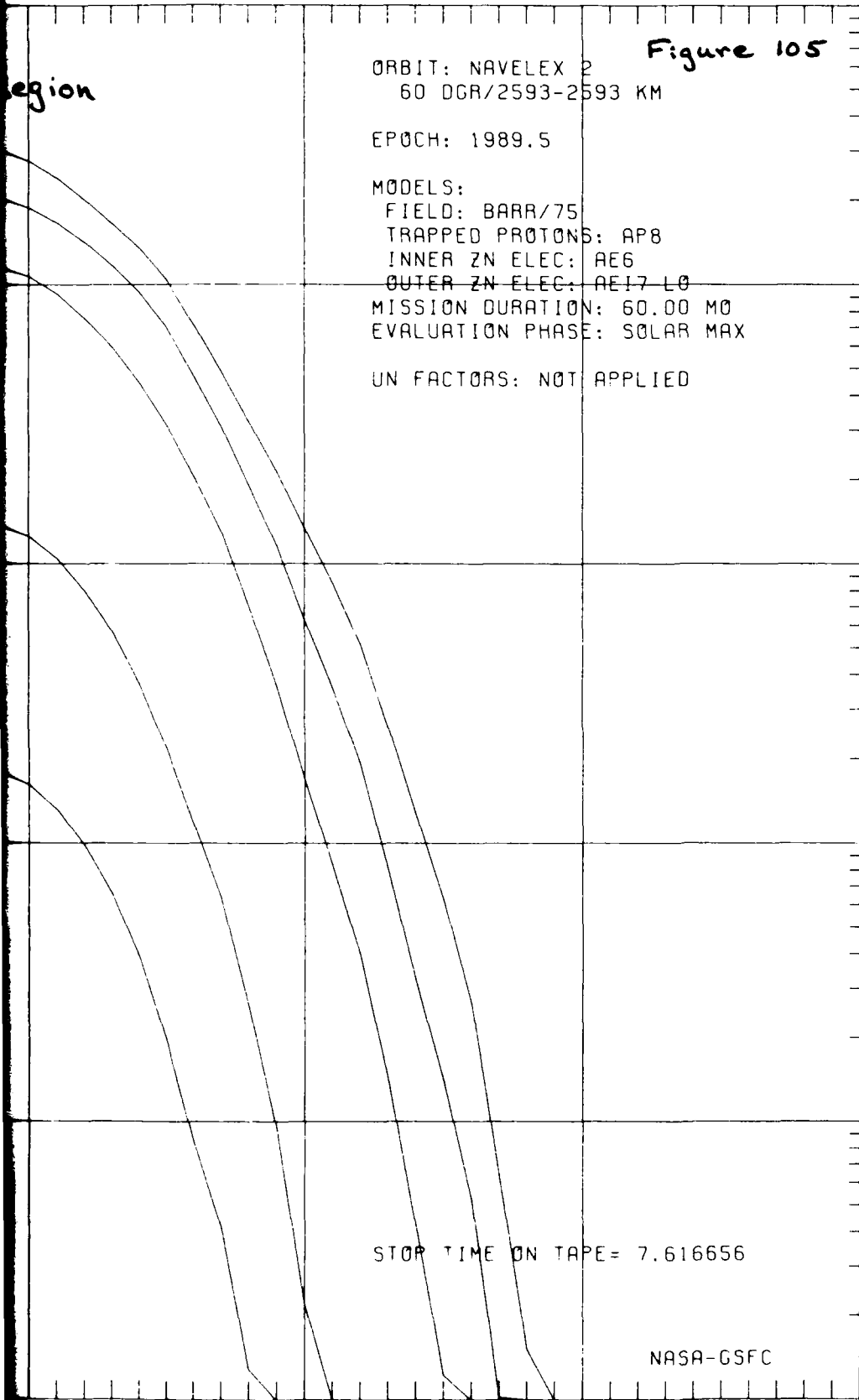
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Region



STOP TIME ON TAPE = 7.616656

NASA-GSFC

10.0 120.0 130.0 140.0

AD-A141 849

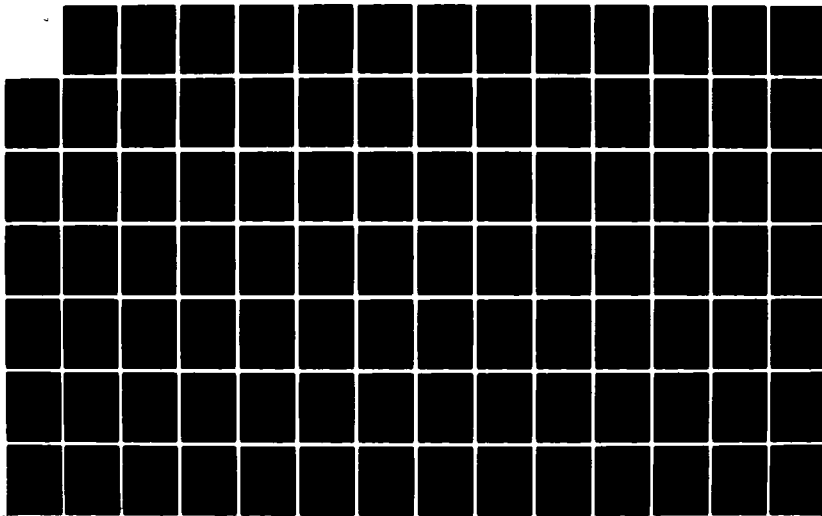
ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR  
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE  
ADMINISTRATION GREENBELT MD GO.. E G STASSINOPOULOS  
NOV 81 NASA-GSFC-X-601-81-28

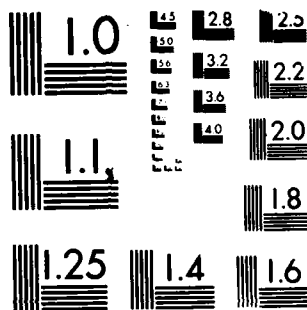
3/5

UNCLASSIFIED

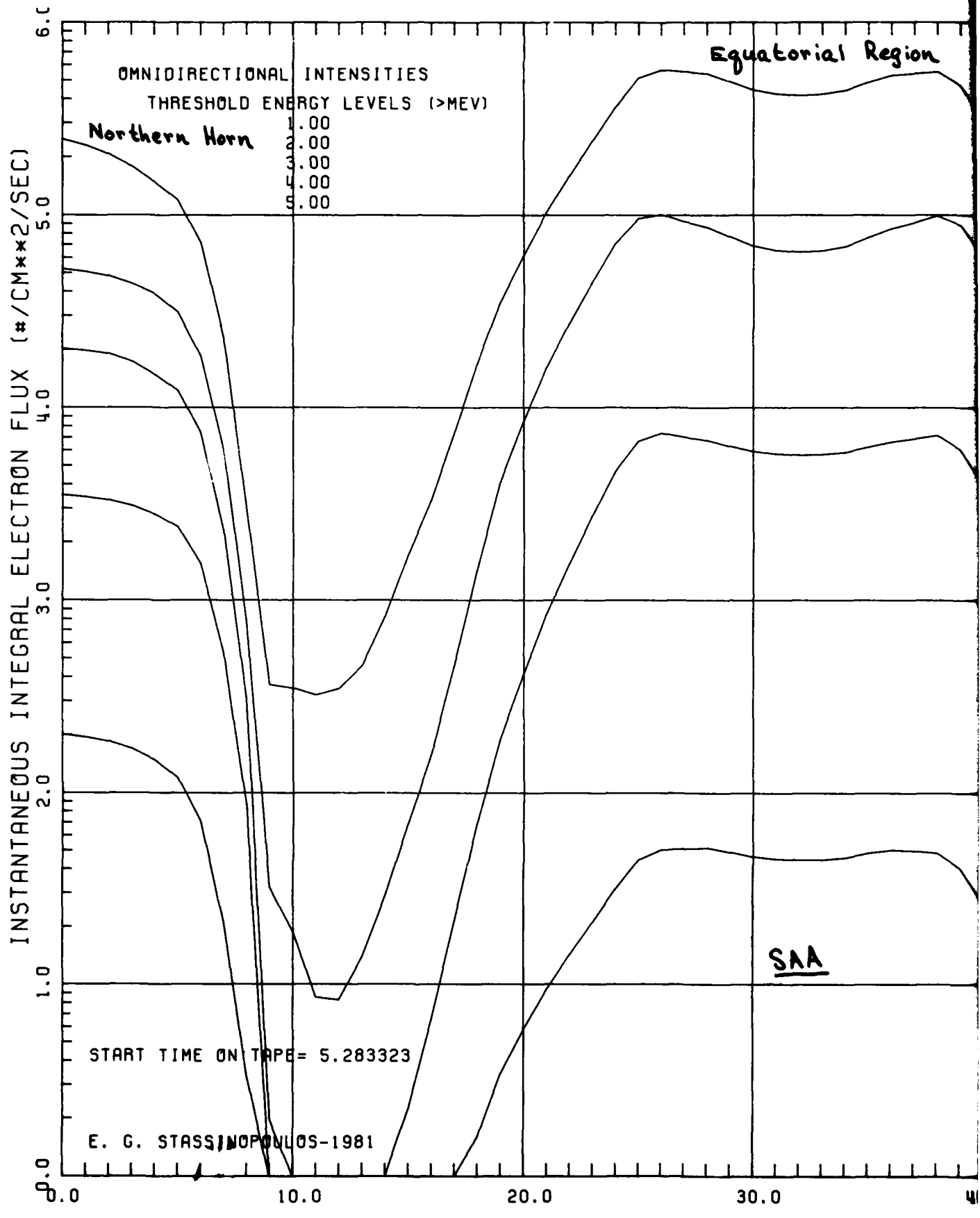
F/G 22/3

NL





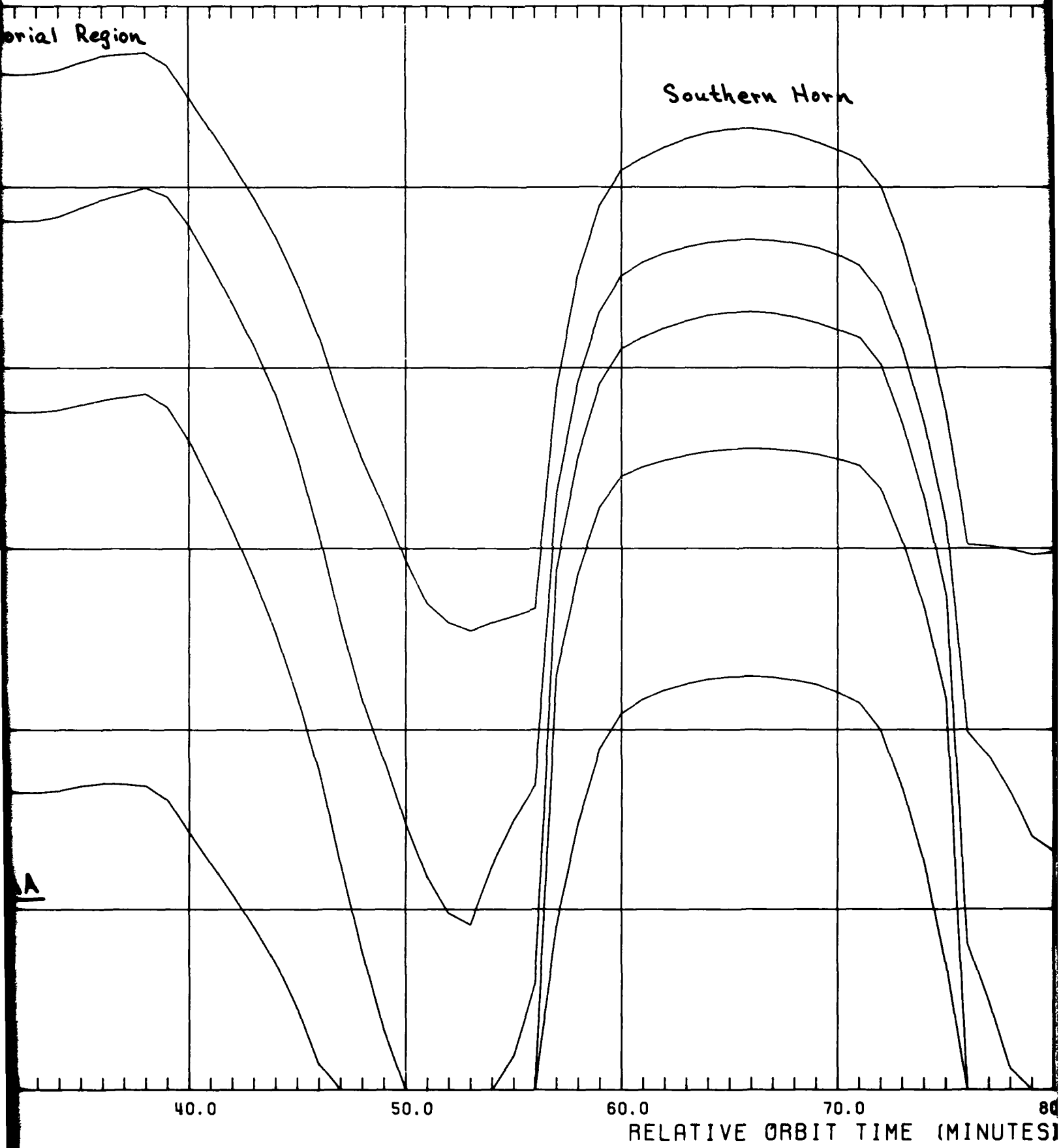
MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A



2

terial Region

Southern Horn

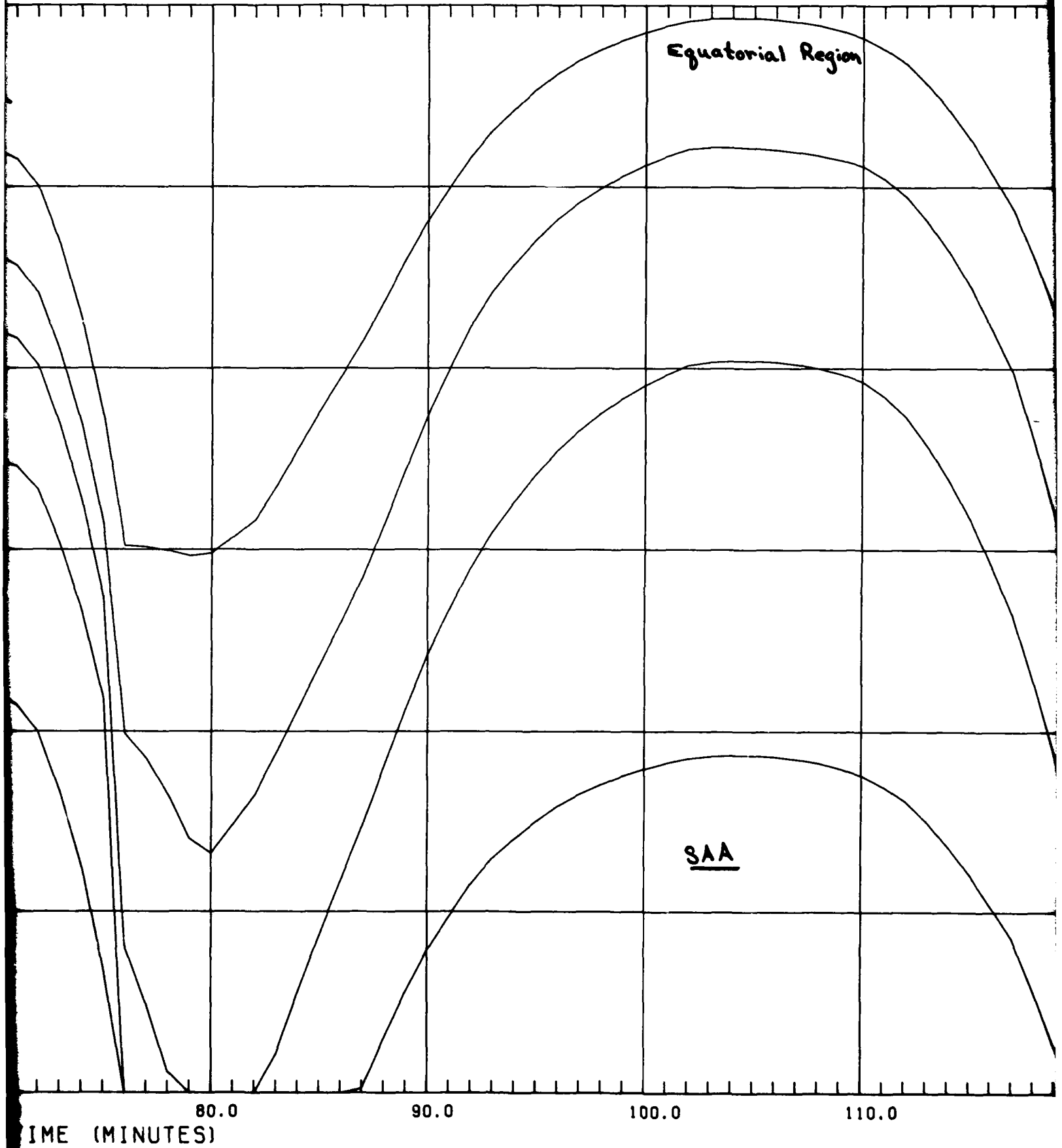


AA

RELATIVE ORBIT TIME (MINUTES)



3



4

Region

Figure 106

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS: Northern Horn

FIELD: BARR/75

TRAPPED PROTONS: AP8

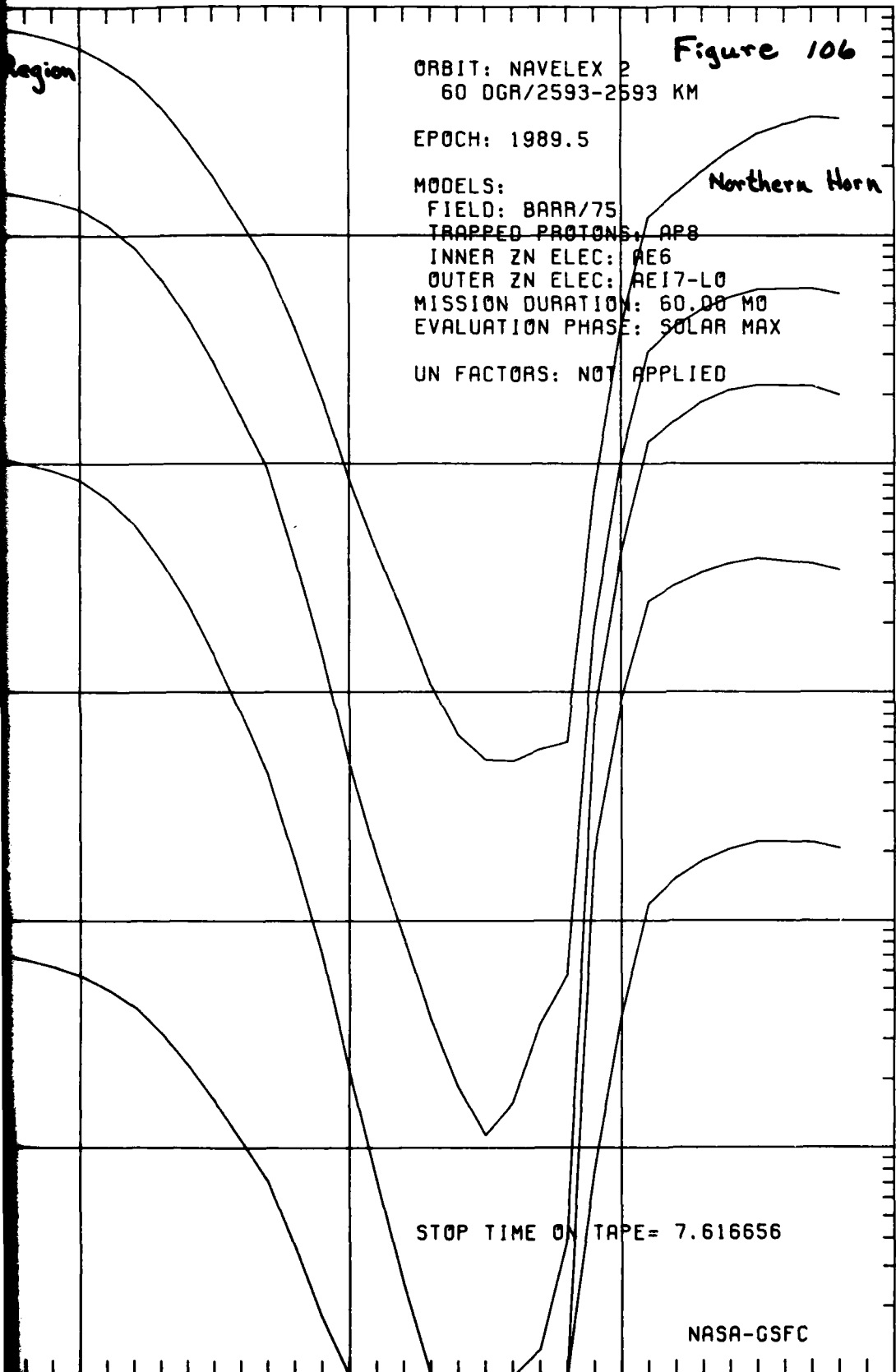
INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED



STOP TIME ON TAPE= 7.616656

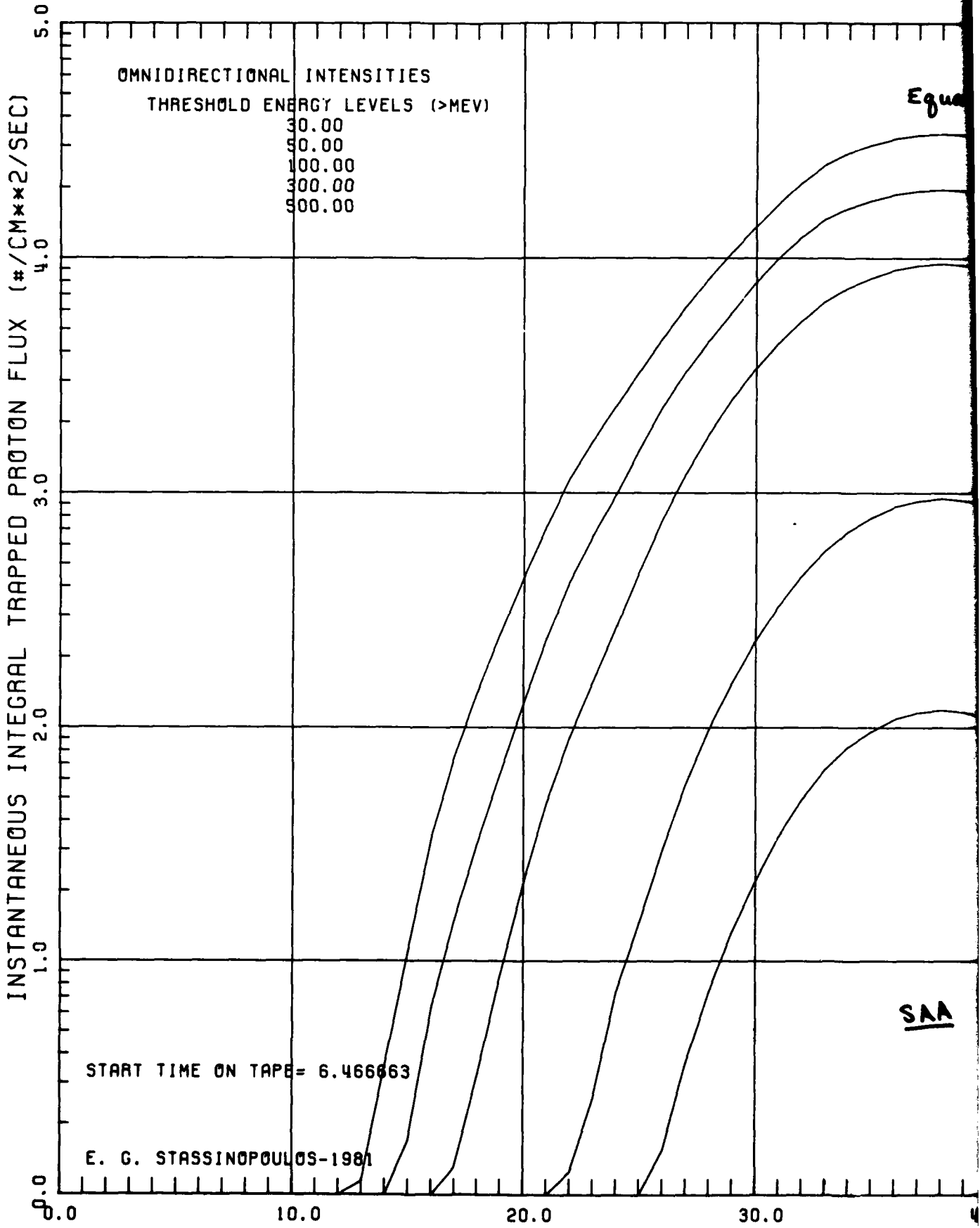
NASA-GSFC

110.0

120.0

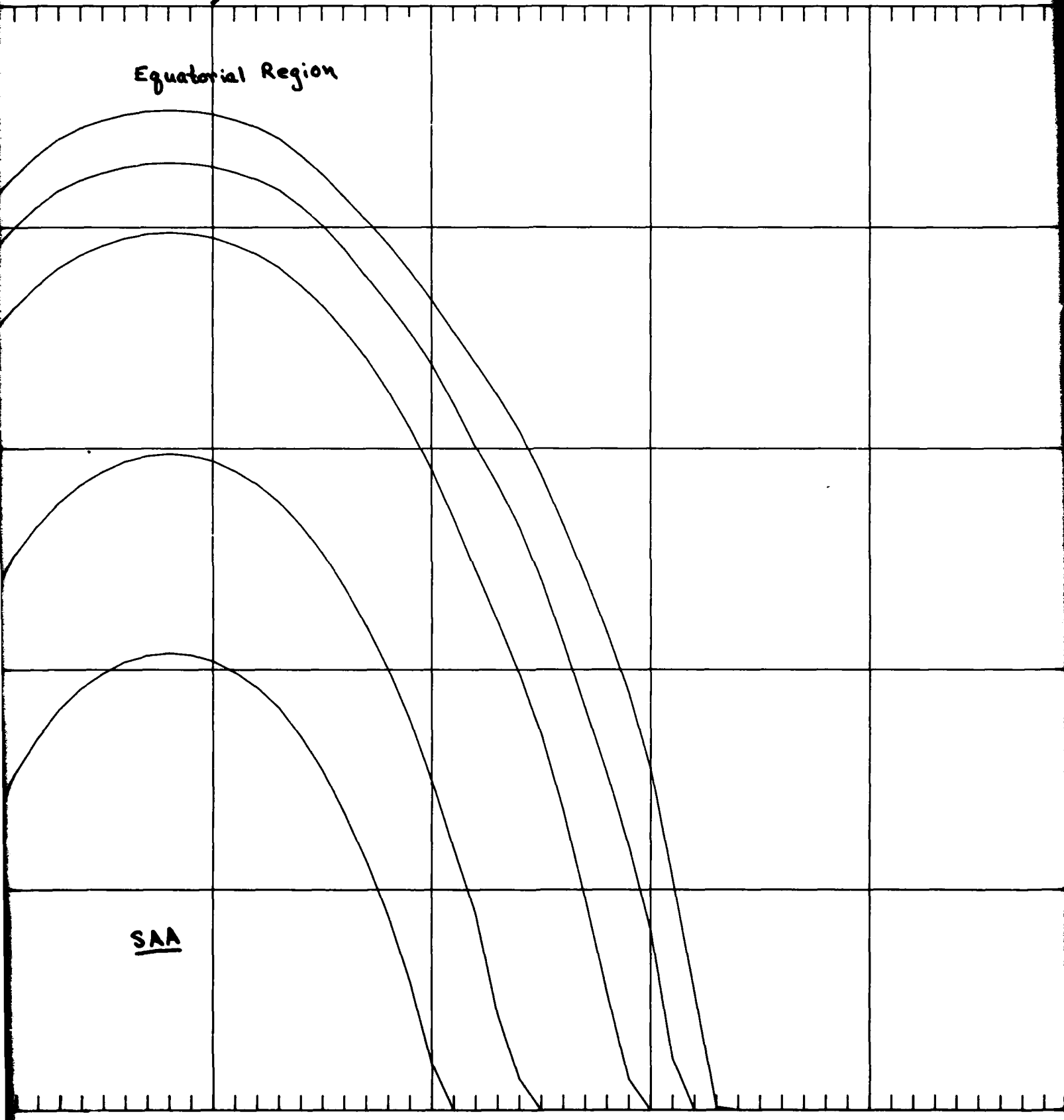
130.0

140.0



1  
2

Equatorial Region



SAA

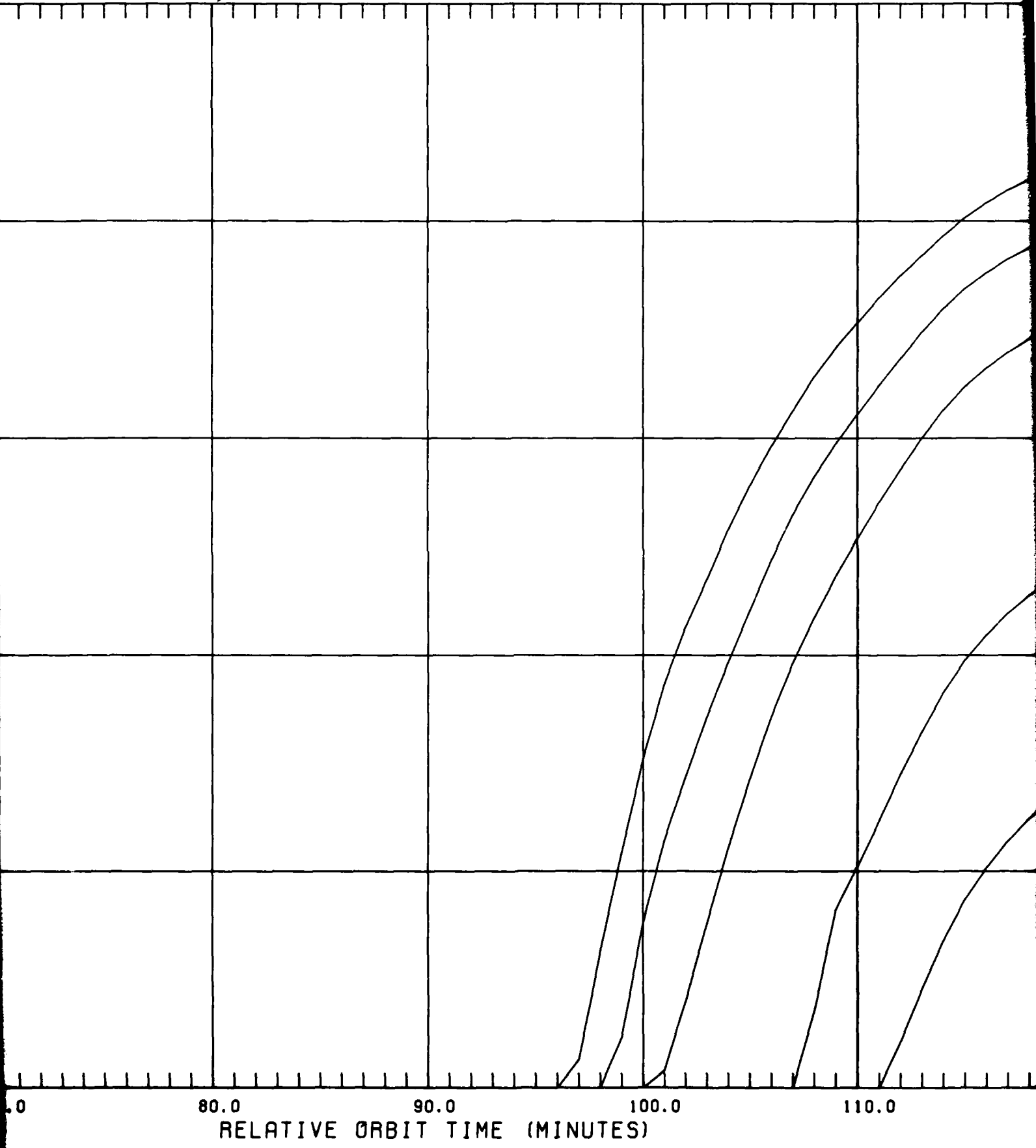
40.0

50.0

60.0

70.0

3



'4

Equatorial Region

SAA

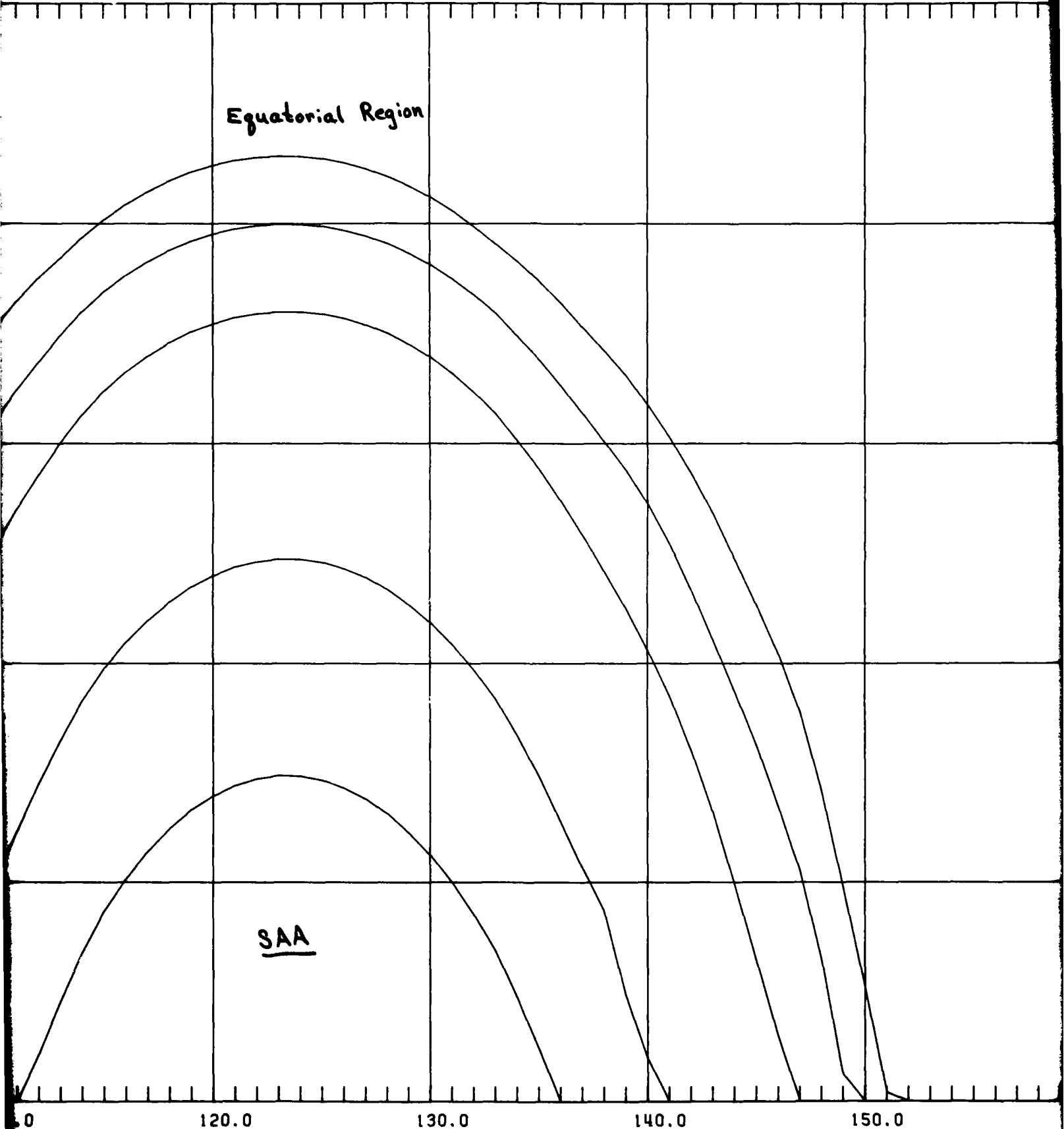


Figure 107

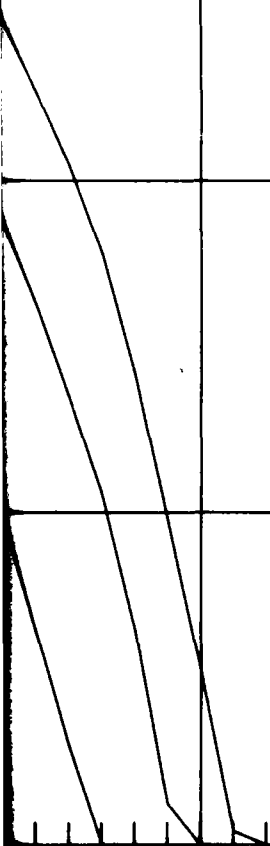
ORBIT: NAVELEX 3  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED



STOP TIME ON TAPE= 9.316662

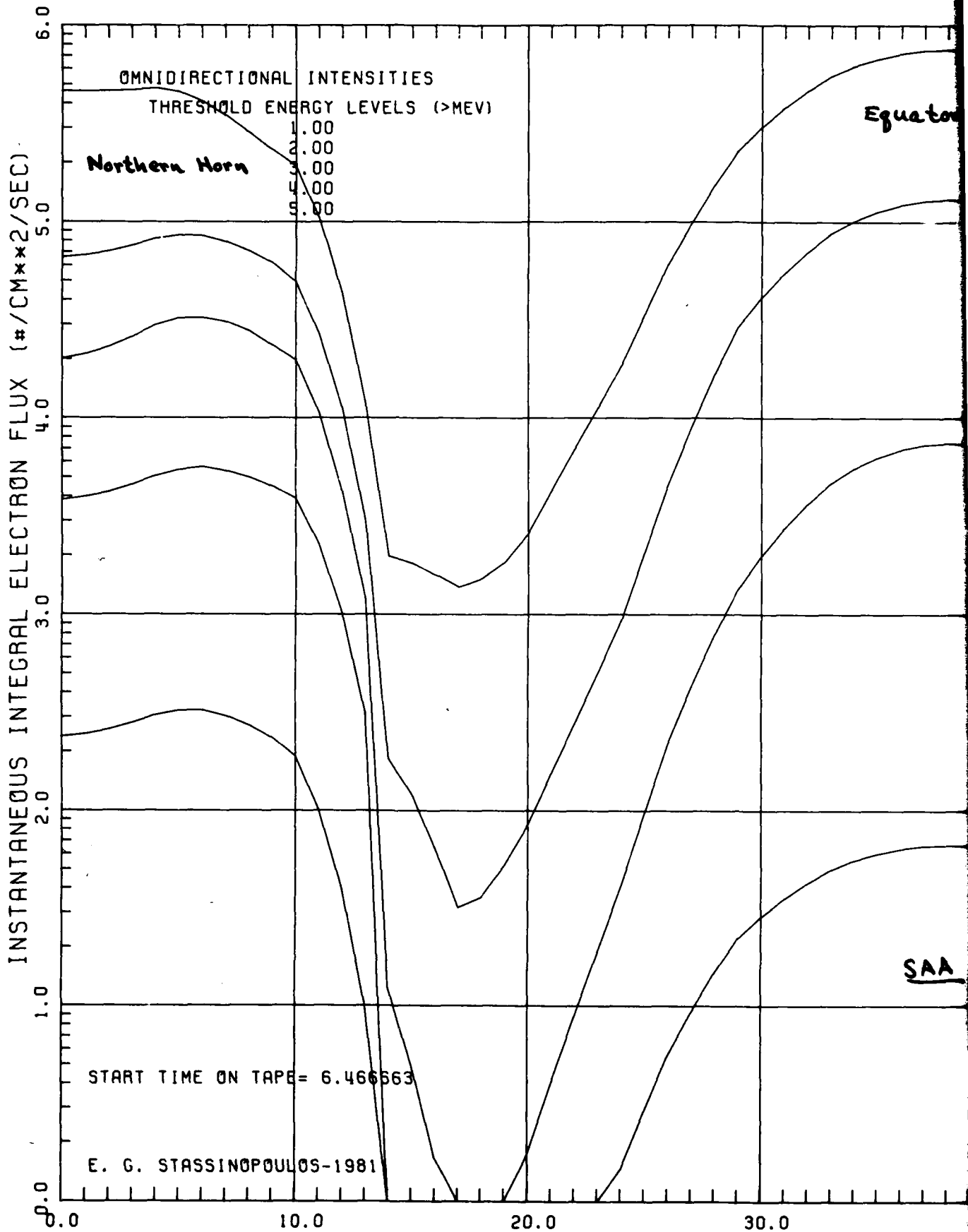
NASA-GSFC

150.0

160.0

170.0

180.0





12

Equatorial Region

SAA

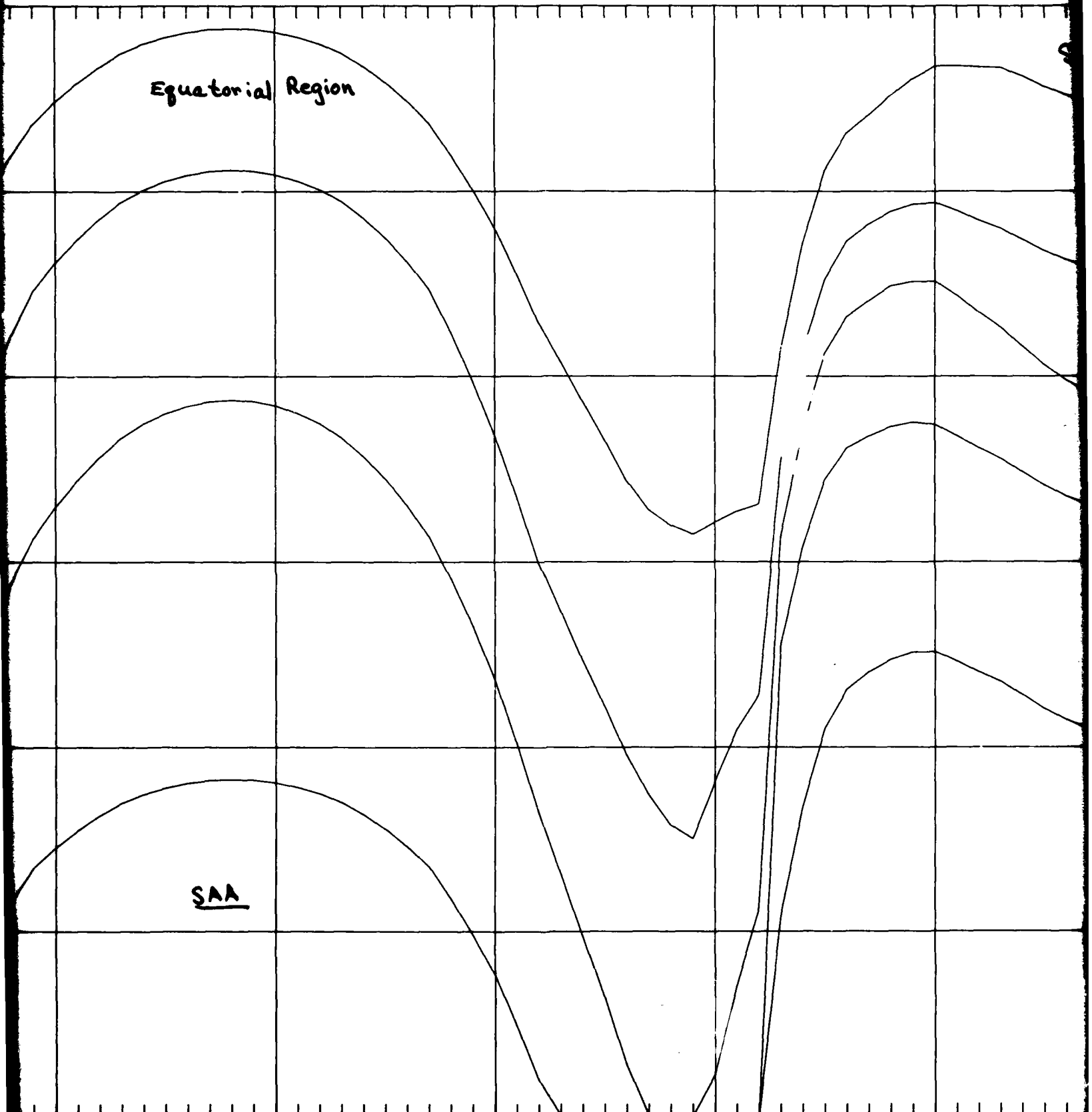
30.0

40.0

50.0

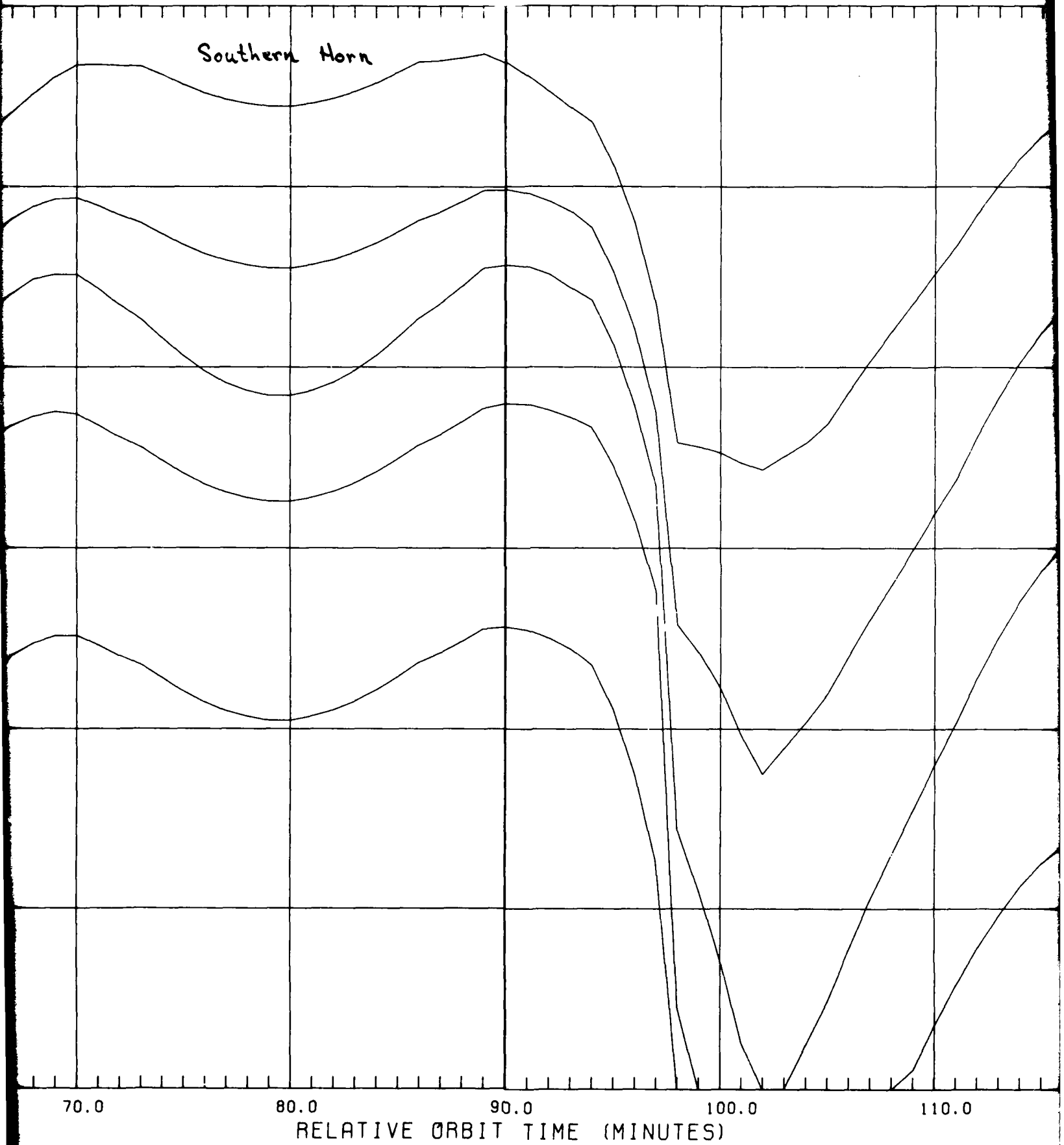
60.0

70.0



13

Southern Horn



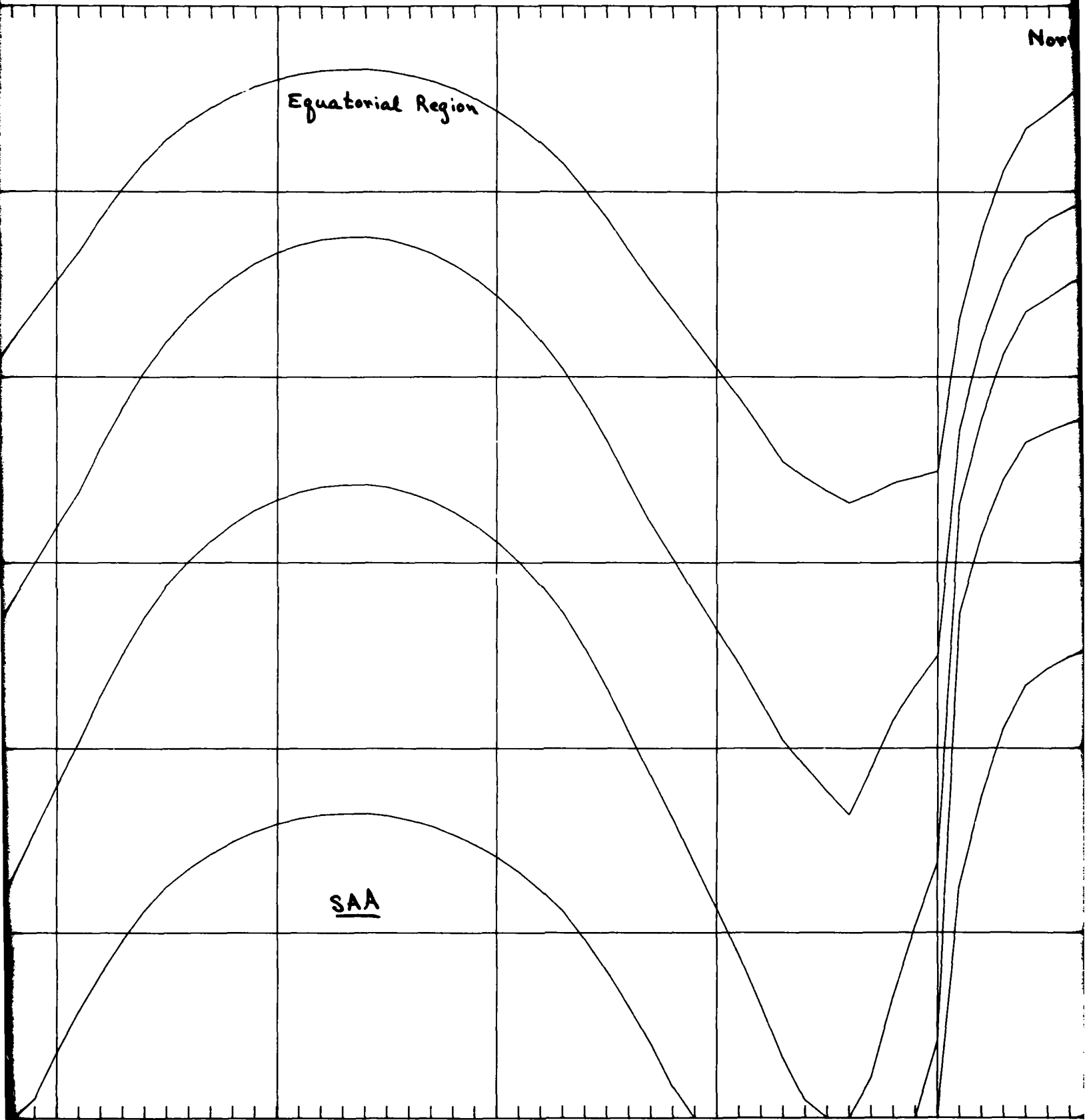
4

Nov

Equatorial Region

SAA

110.0 120.0 130.0 140.0 150.0



5

Northern Horn

Figure 108

ORBIT: NAVELEX B  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 9.316662

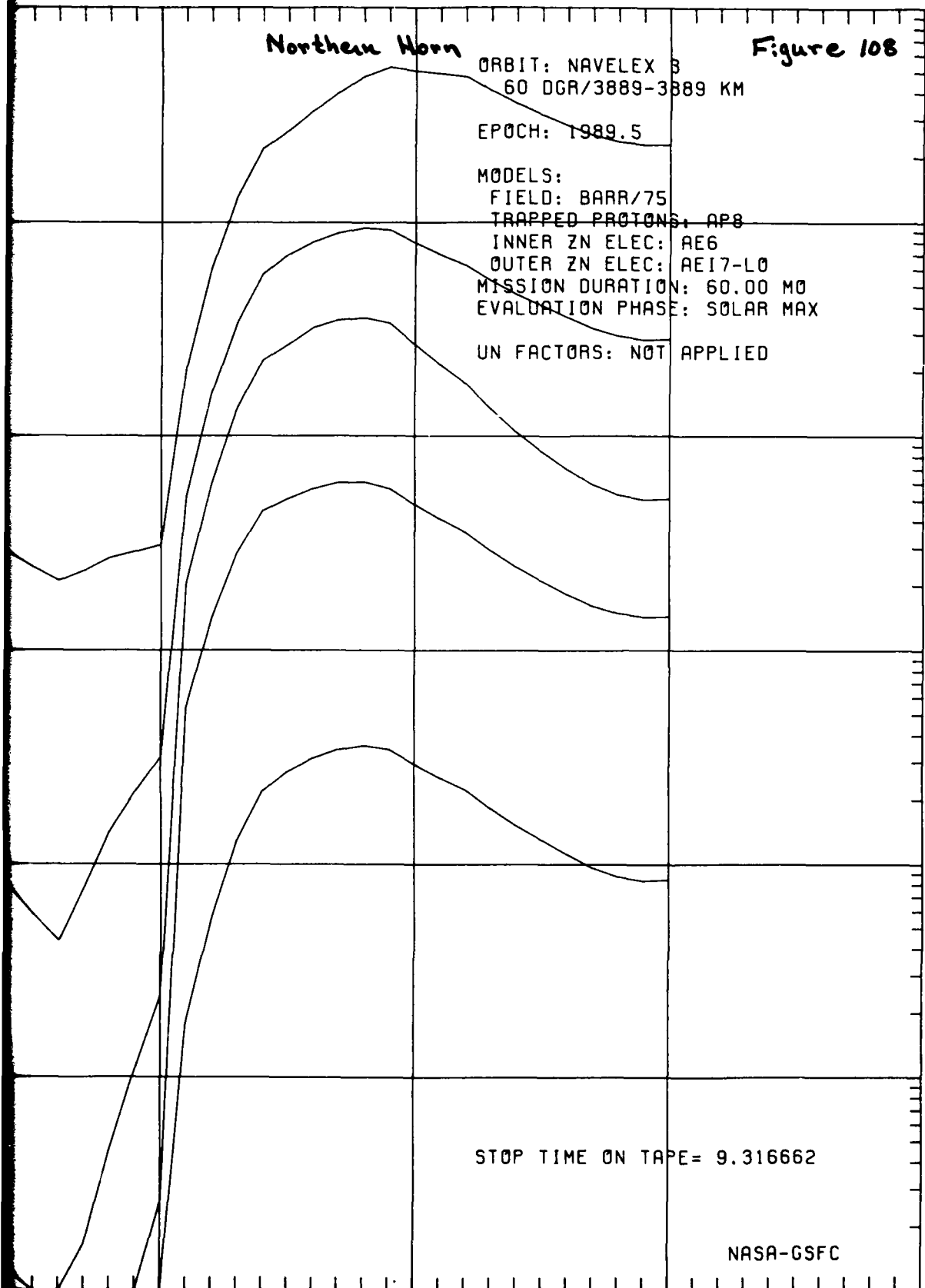
NASA-GSFC

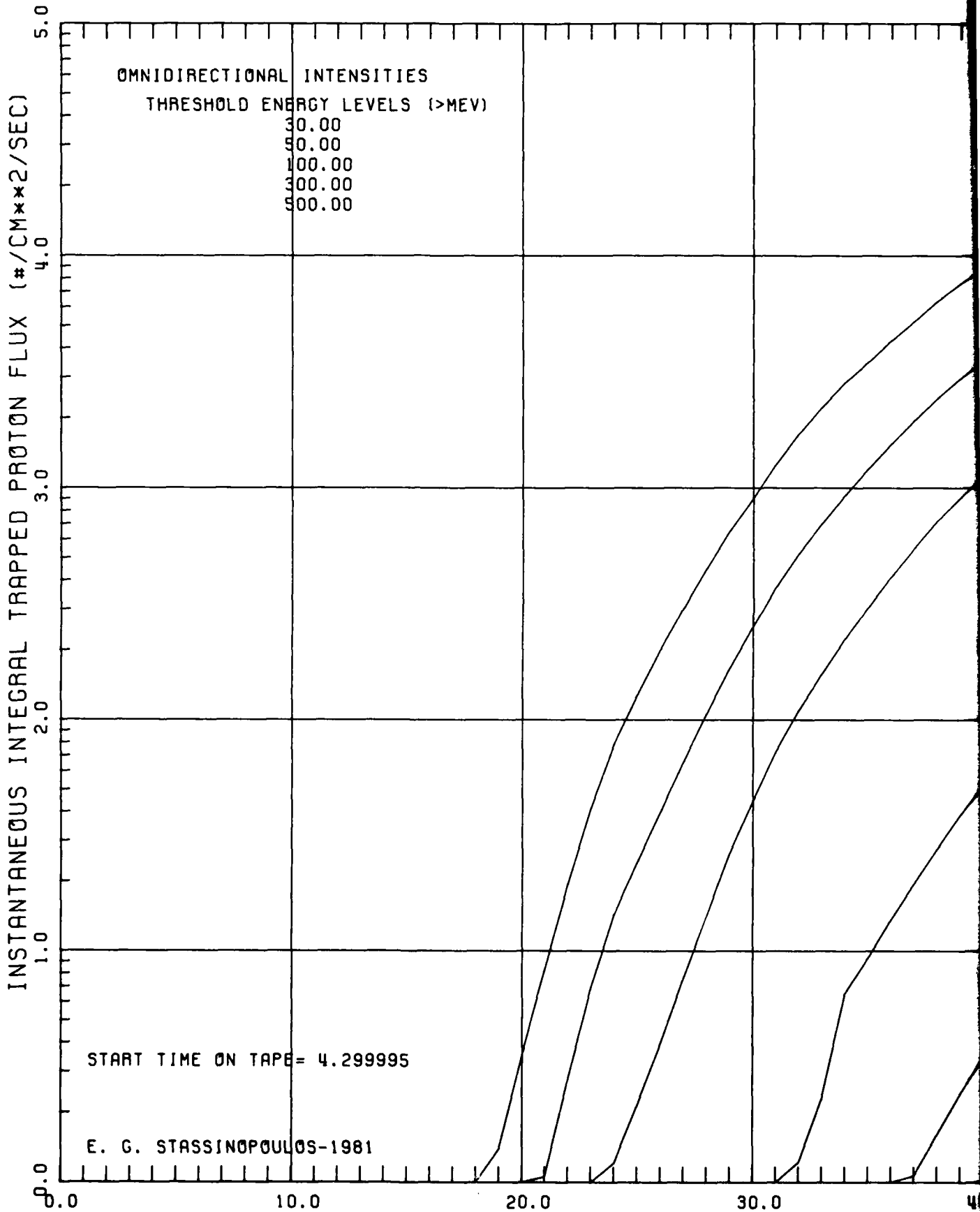
150.0

160.0

170.0

180.0





1  
2

Equatorial Region

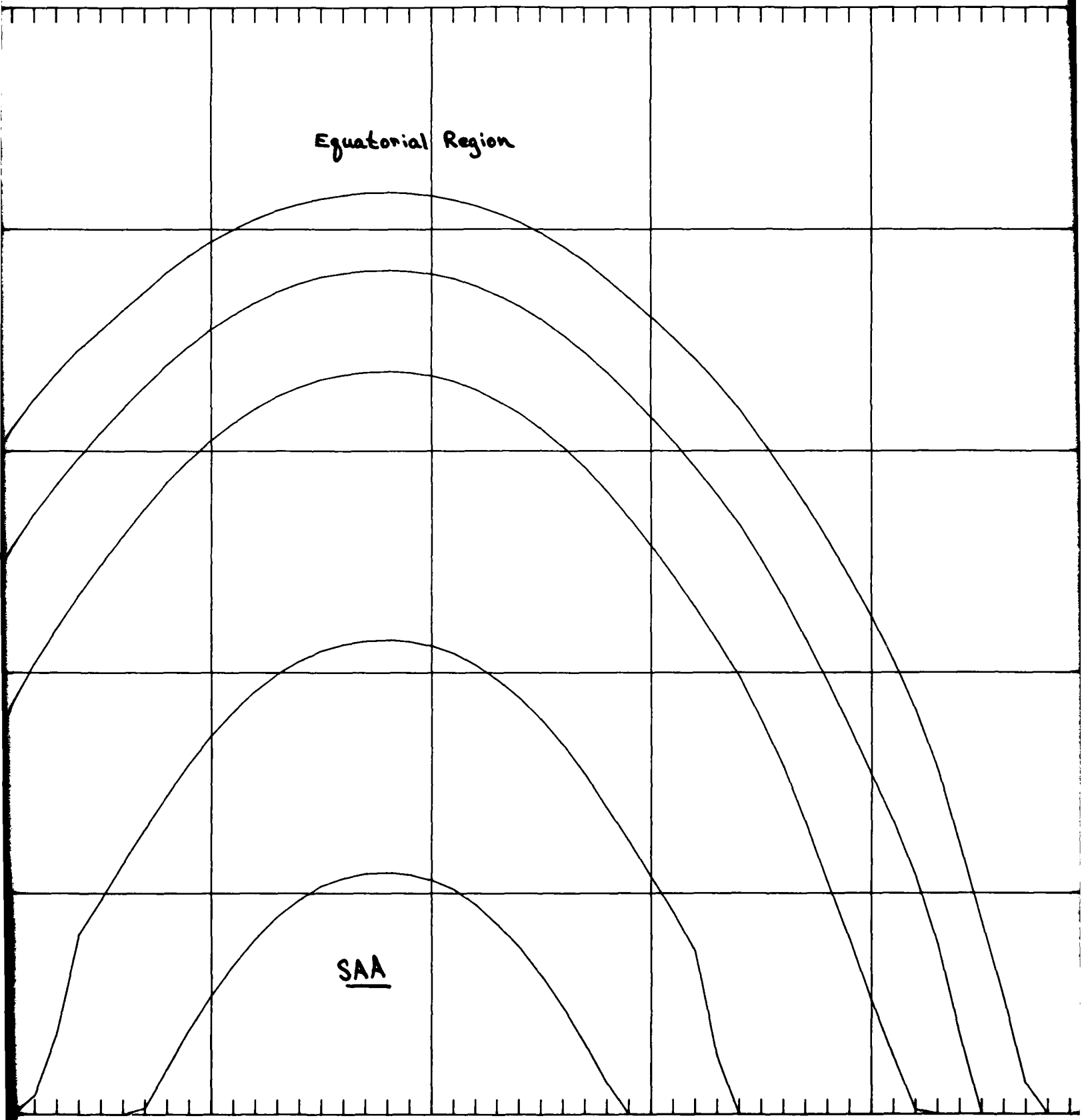
SAA

40.0

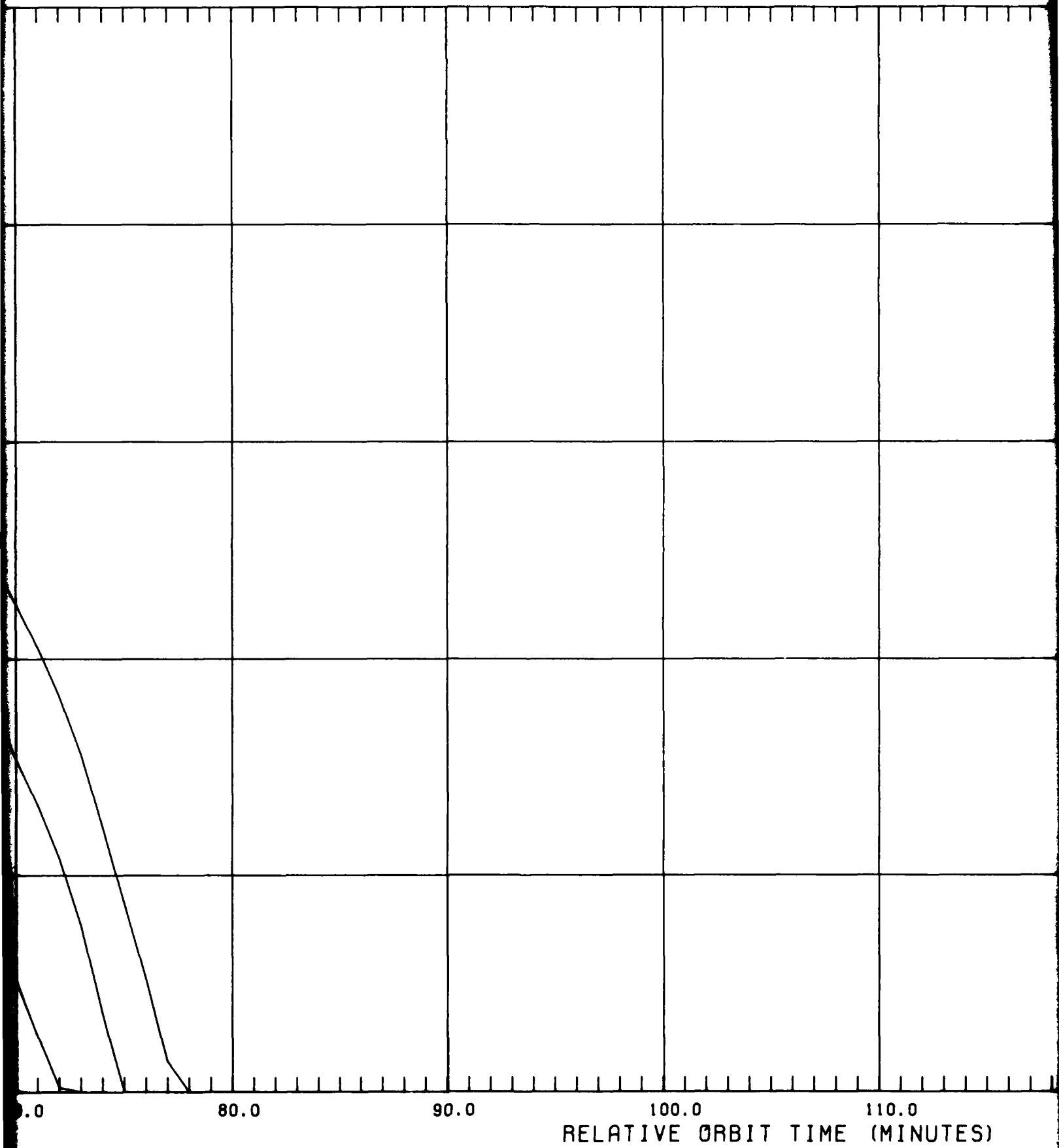
50.0

60.0

70.0



3'



RELATIVE ORBIT TIME (MINUTES)

4'

Equatorial

SAA

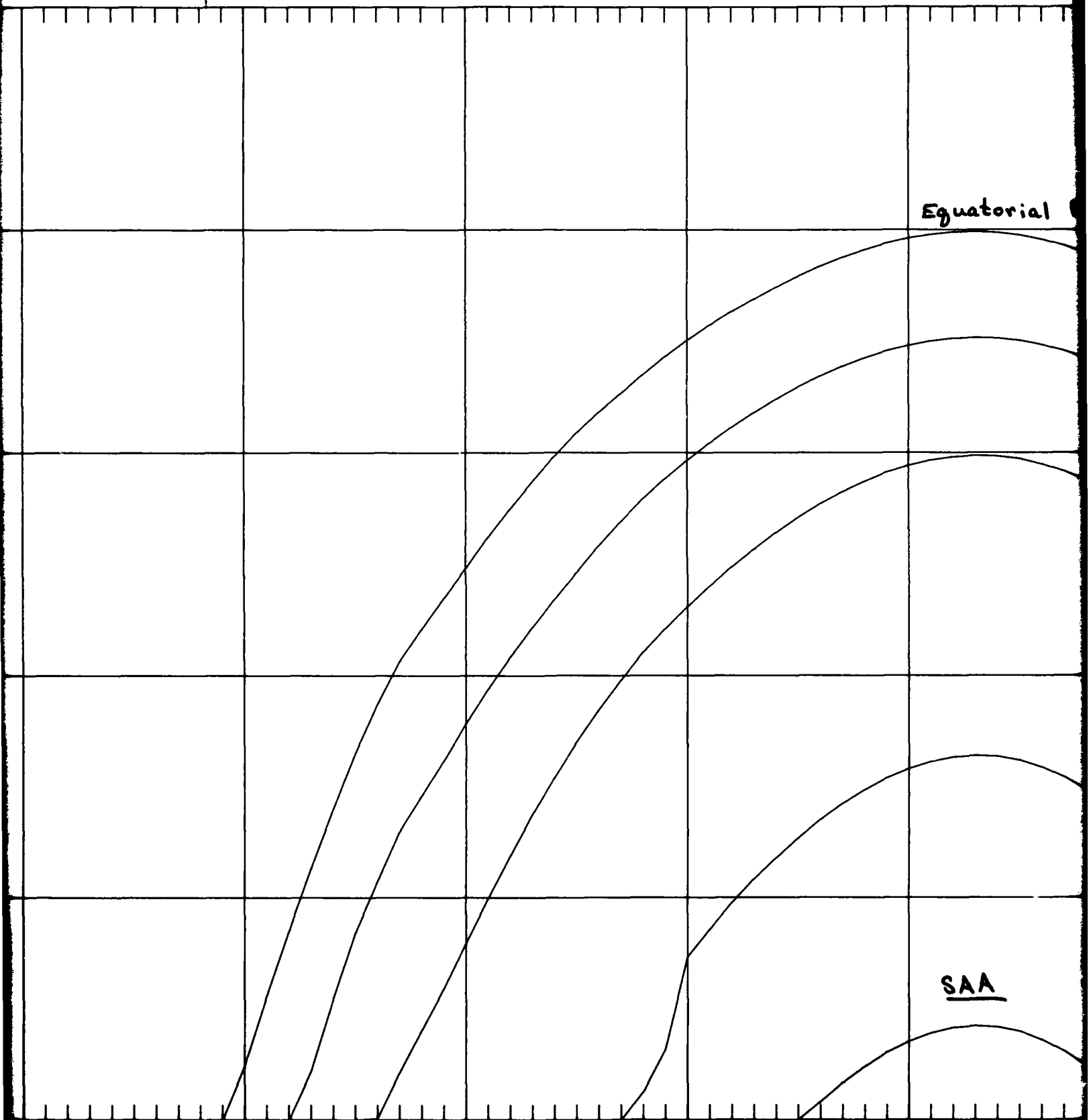
10.0  
(MINUTES)

120.0

130.0

140.0

150.0





5.

Equatorial Region

ORBIT: NA  
60 DGR/S  
EPOCH: 19  
MODELS:  
FIELD: BA  
TRAPPED P  
INNER ZN  
OUTER ZN  
MISSION DU  
EVALUATION  
UN FACTORS

SAA

STOP TIME

50.0 160.0 170.0 180.0 190.0

Figure 109

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

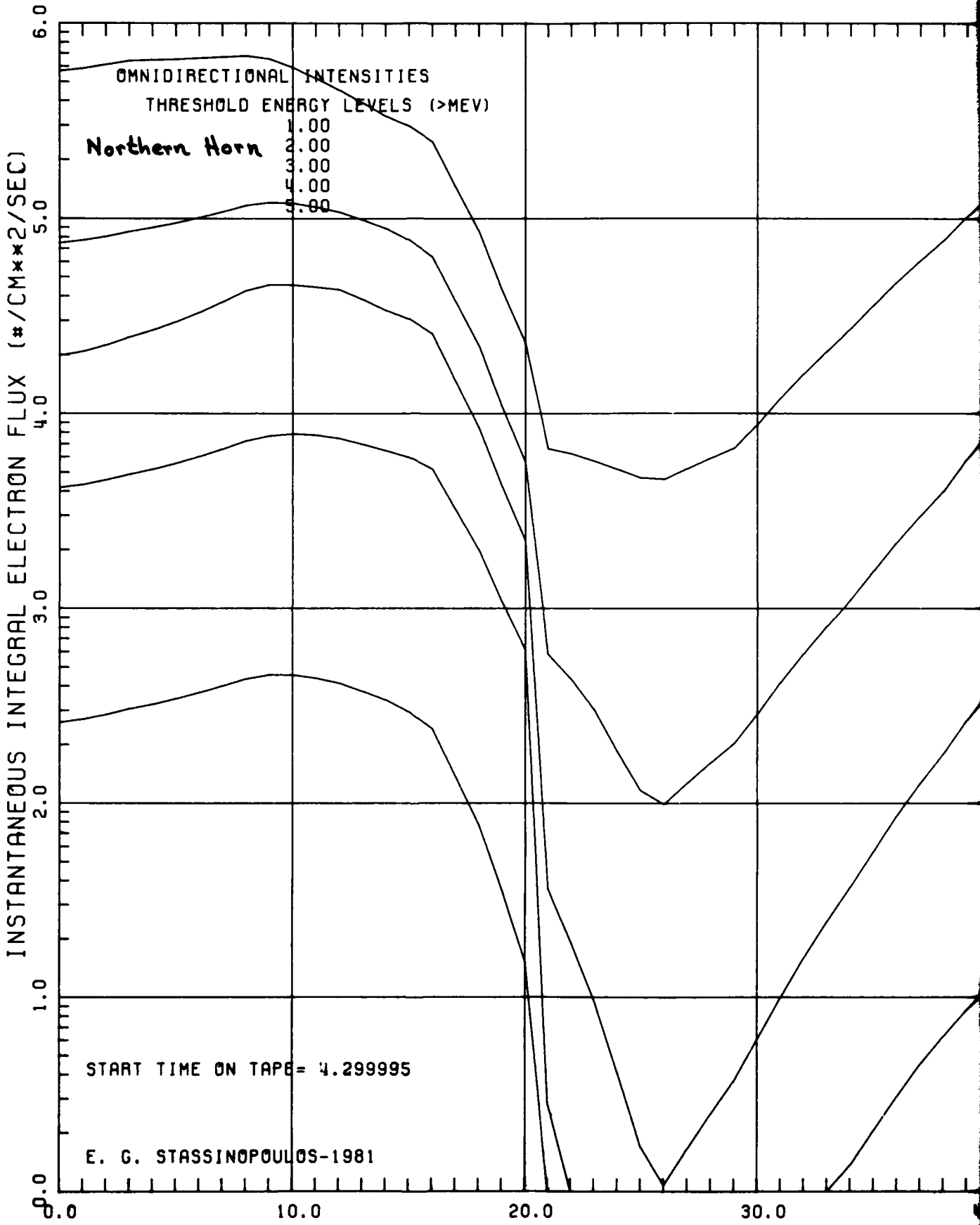
STOP TIME ON TAPE= 7.716662

NASA-GSFC

190.0

200.0

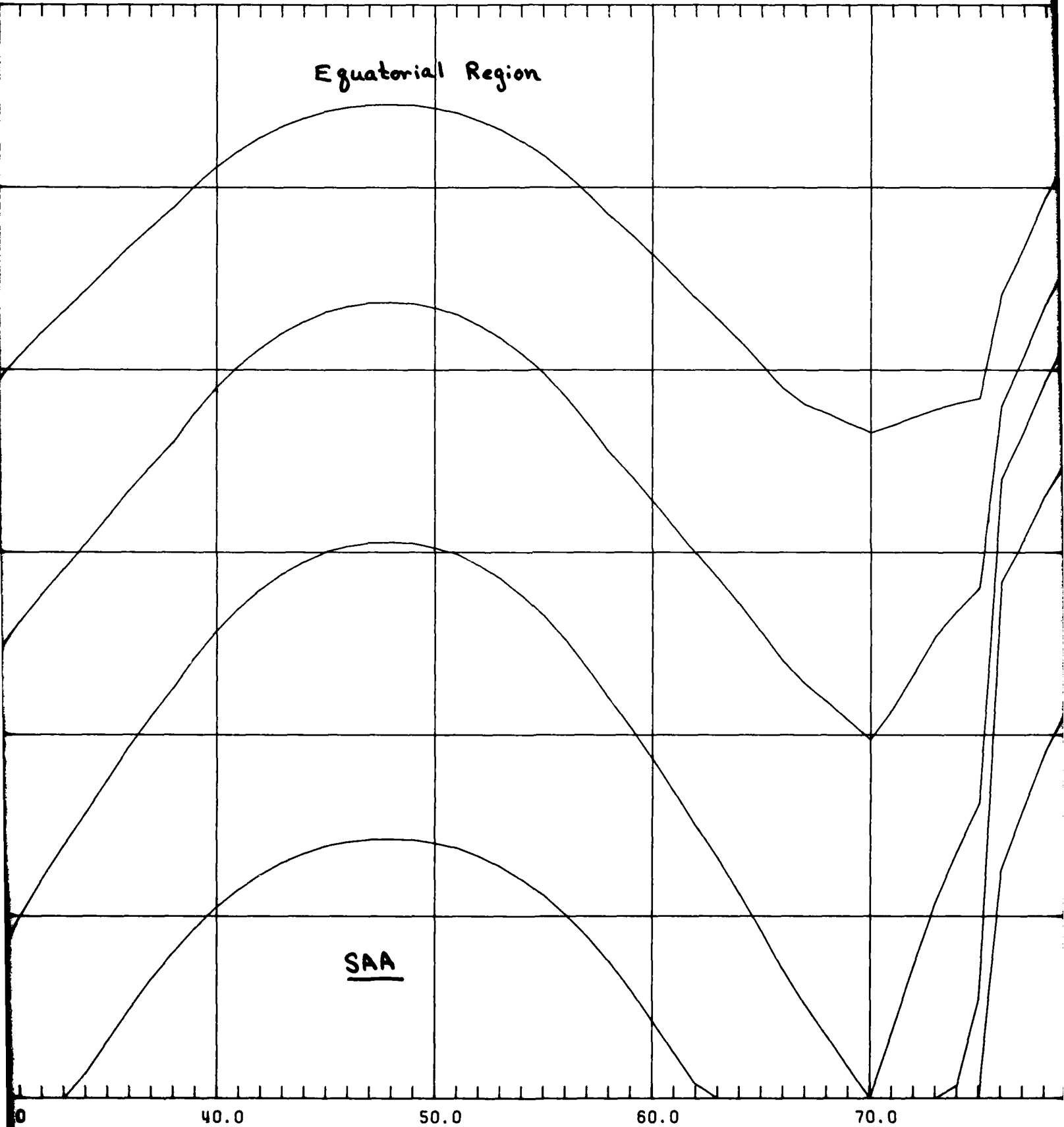
210.0



2

Equatorial Region

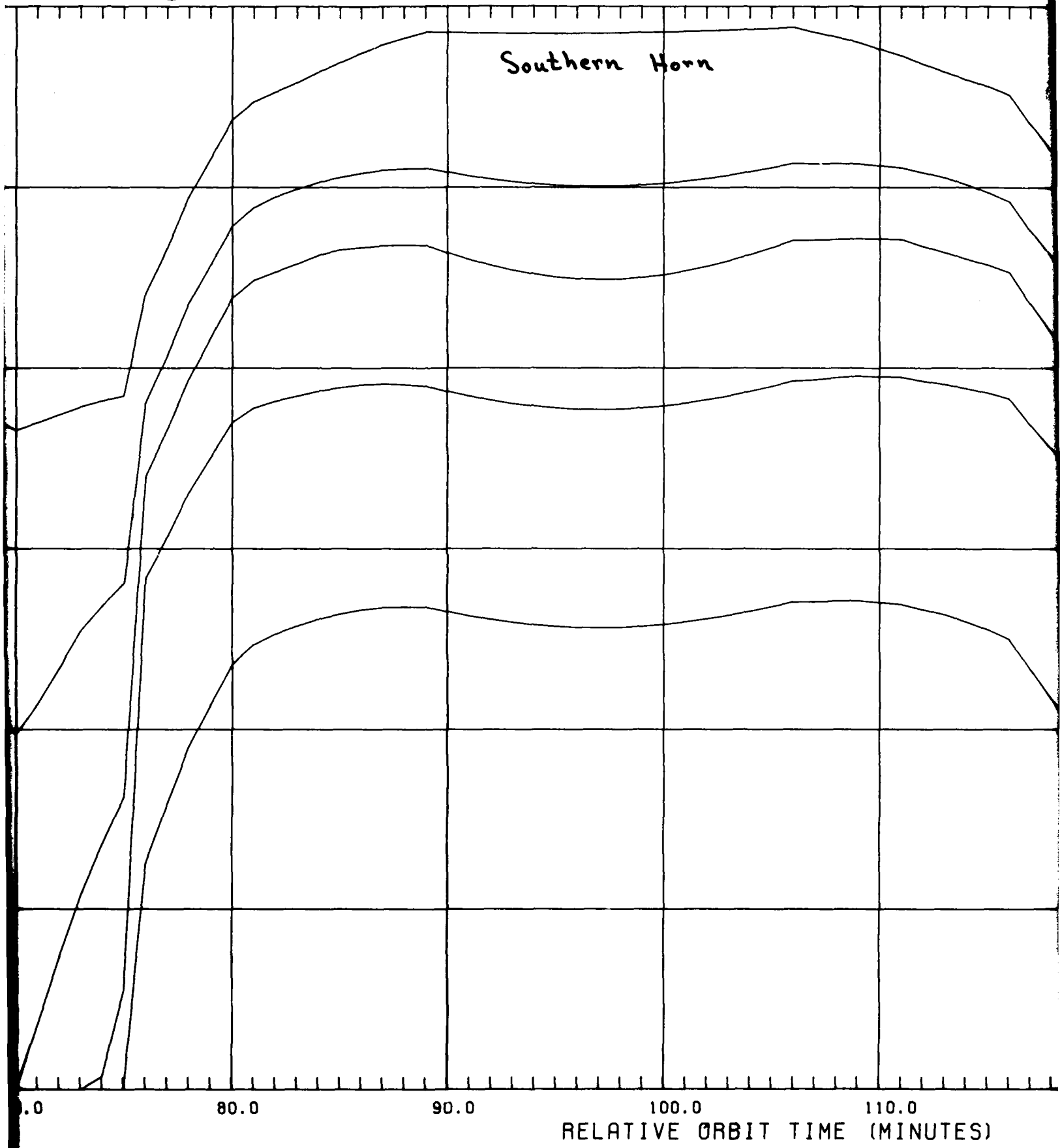
SAA



1

3'

Southern Horn



U

Equatorial

SAA

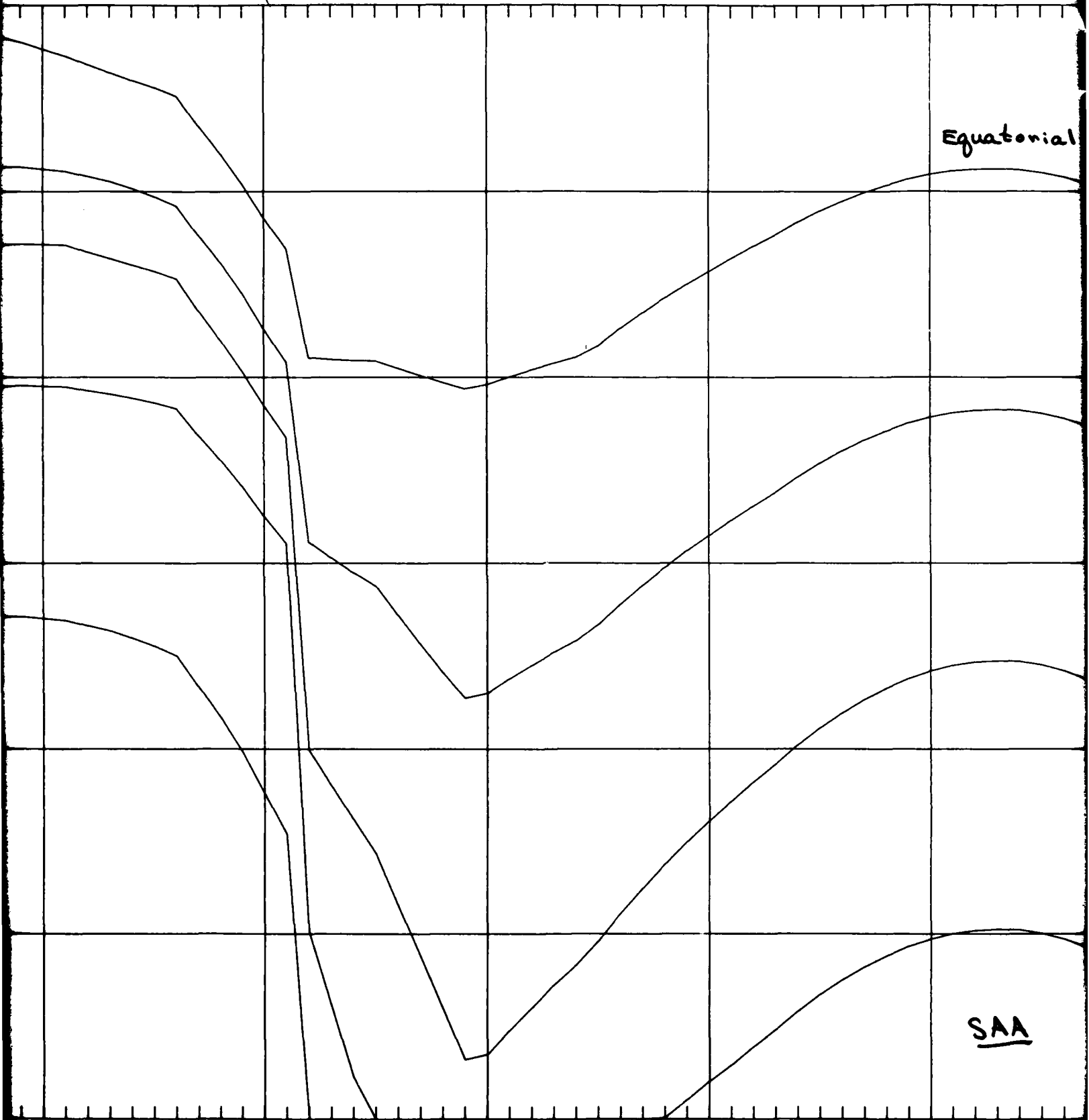
110.0  
(MINUTES)

120.0

130.0

140.0

150.0



5'

Equatorial Region

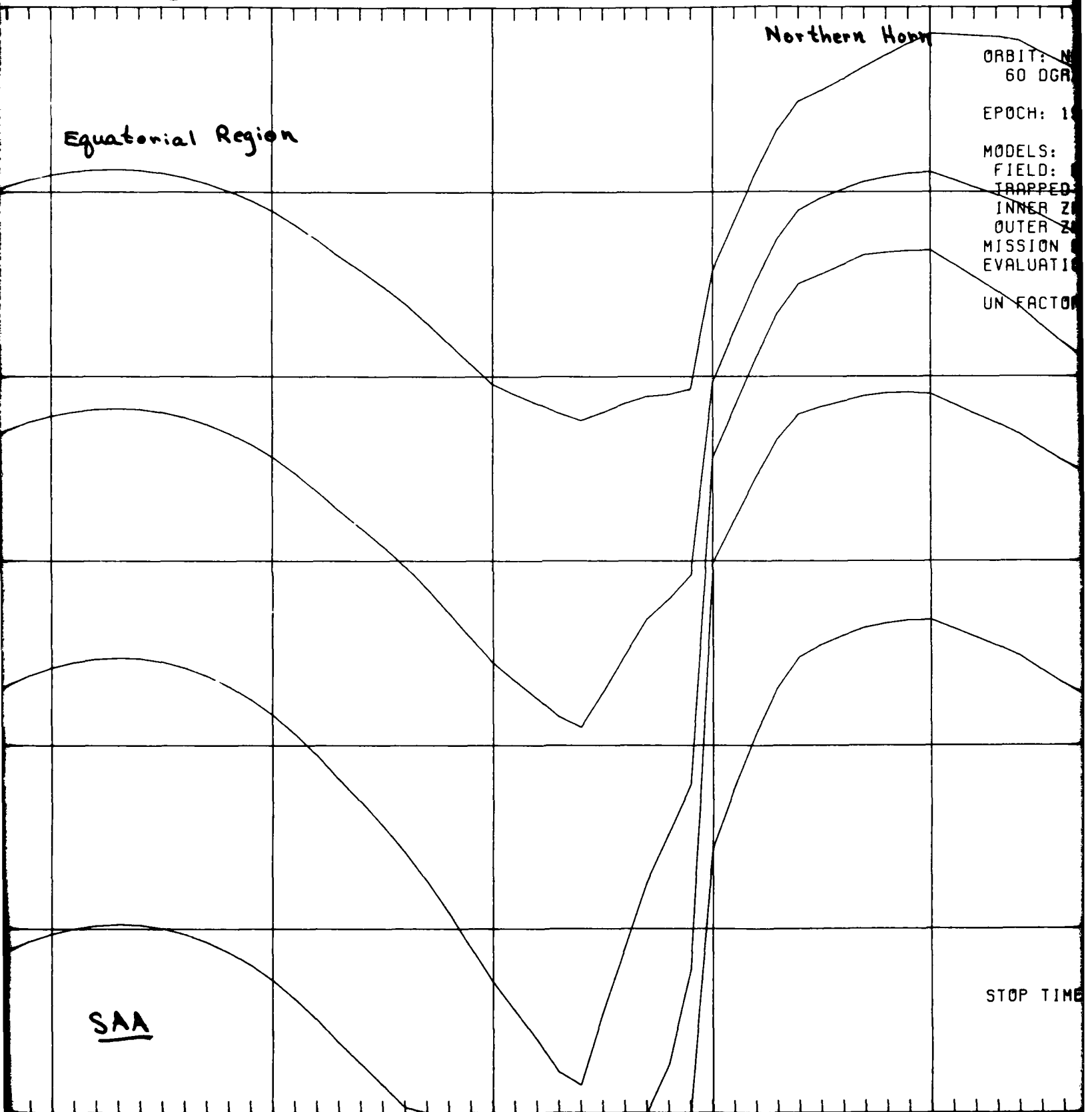
Northern Hemisphere

ORBIT: N  
60 DGR  
EPOCH: 1  
MODELS:  
FIELD:  
TRAPPED  
INNER Z  
OUTER Z  
MISSION  
EVALUATI  
UN FACTOR

SAA

STOP TIME

150.0 160.0 170.0 180.0 190.0



Northern Hemisphere

Figure 110

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: RE6

OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 7.716662

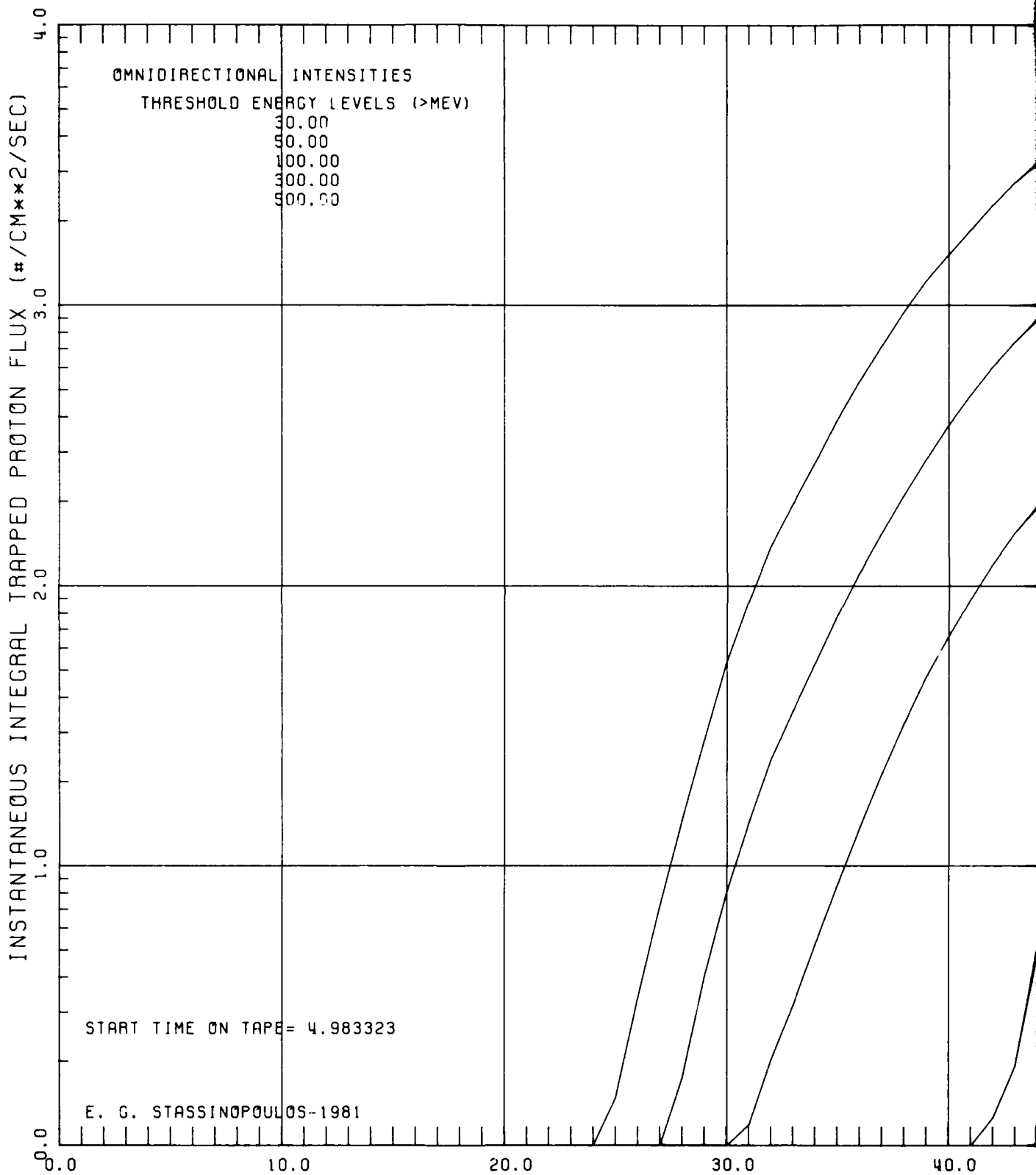
NASA-GSFC

190.0

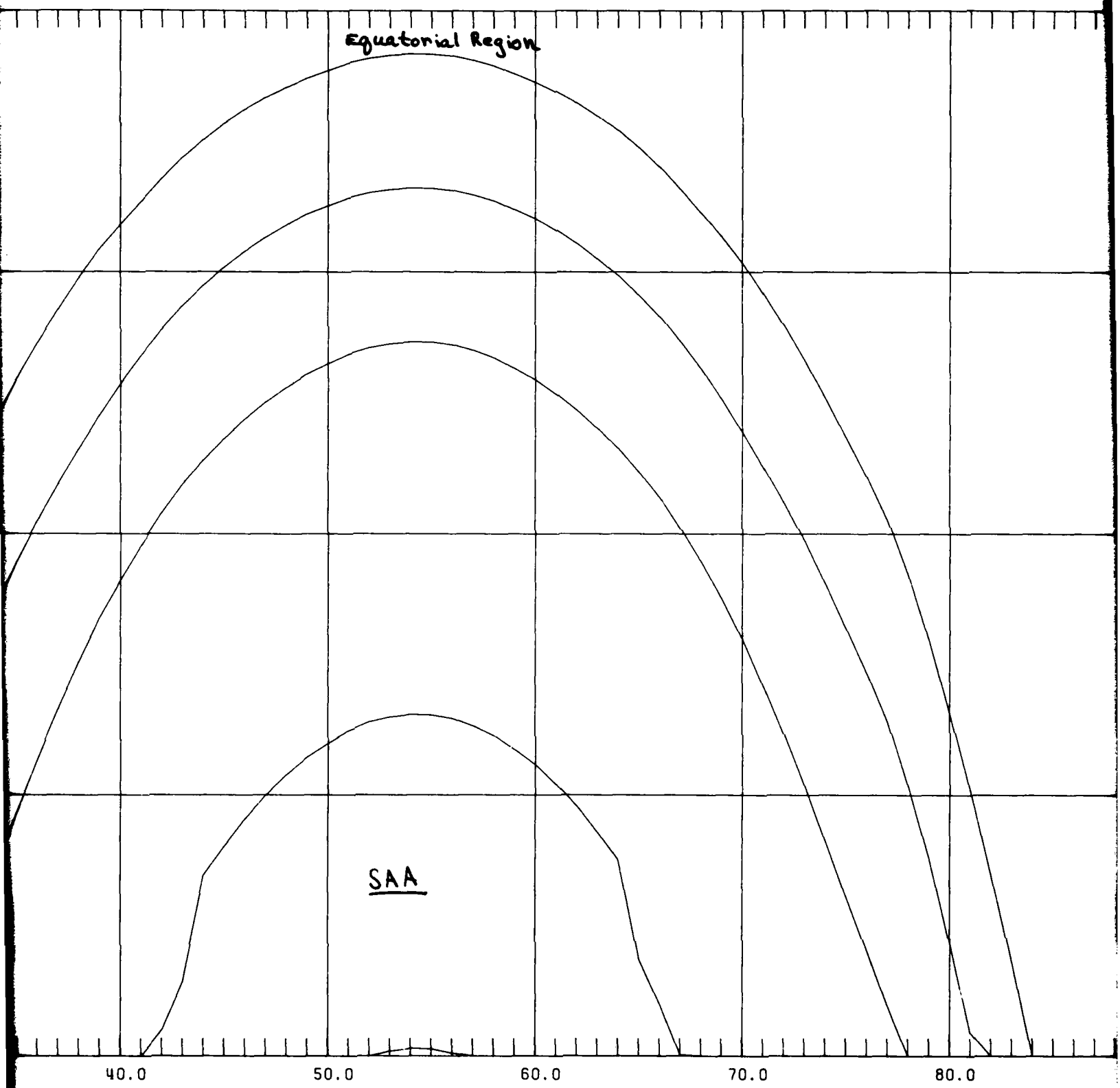
200.0

210.0

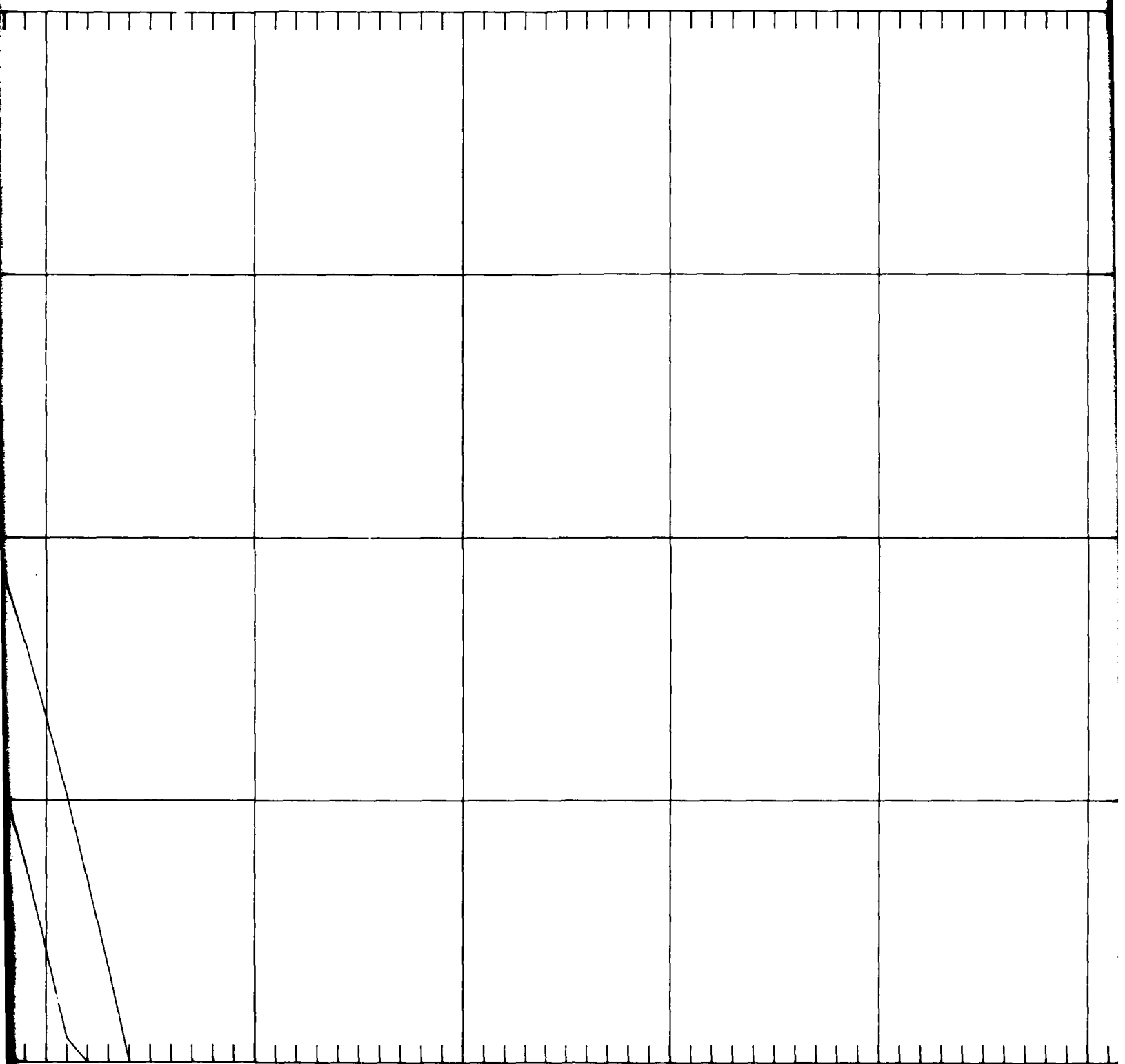




2'



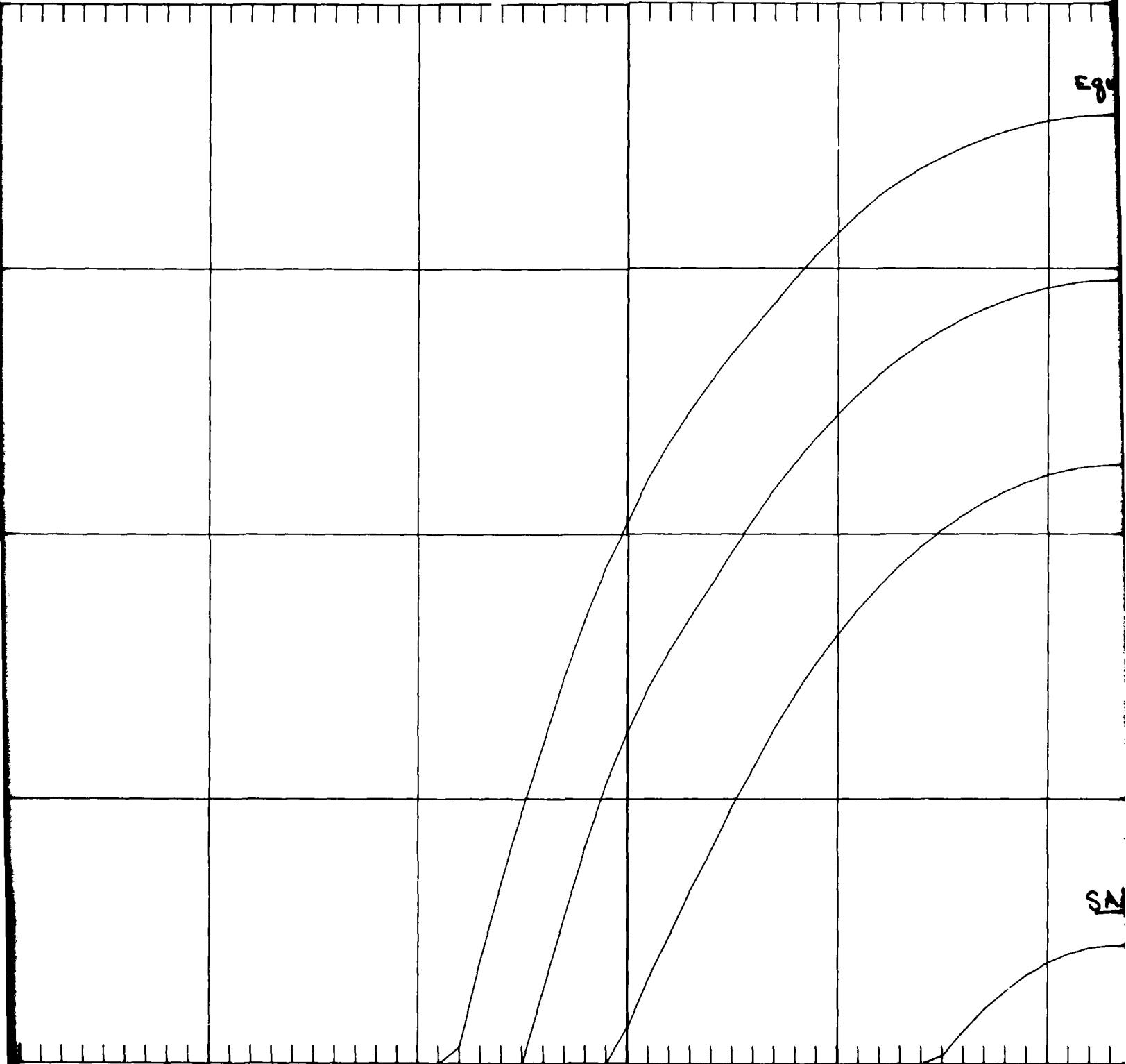
3'



80.0 90.0 100.0 110.0 120.0 130.0

RELATIVE ORBIT TIME (MINUTES)

4



0 130.0 140.0 150.0 160.0 170.0  
TIME (MINUTES)

Eg

SN

5'

Equatorial Region

SAA

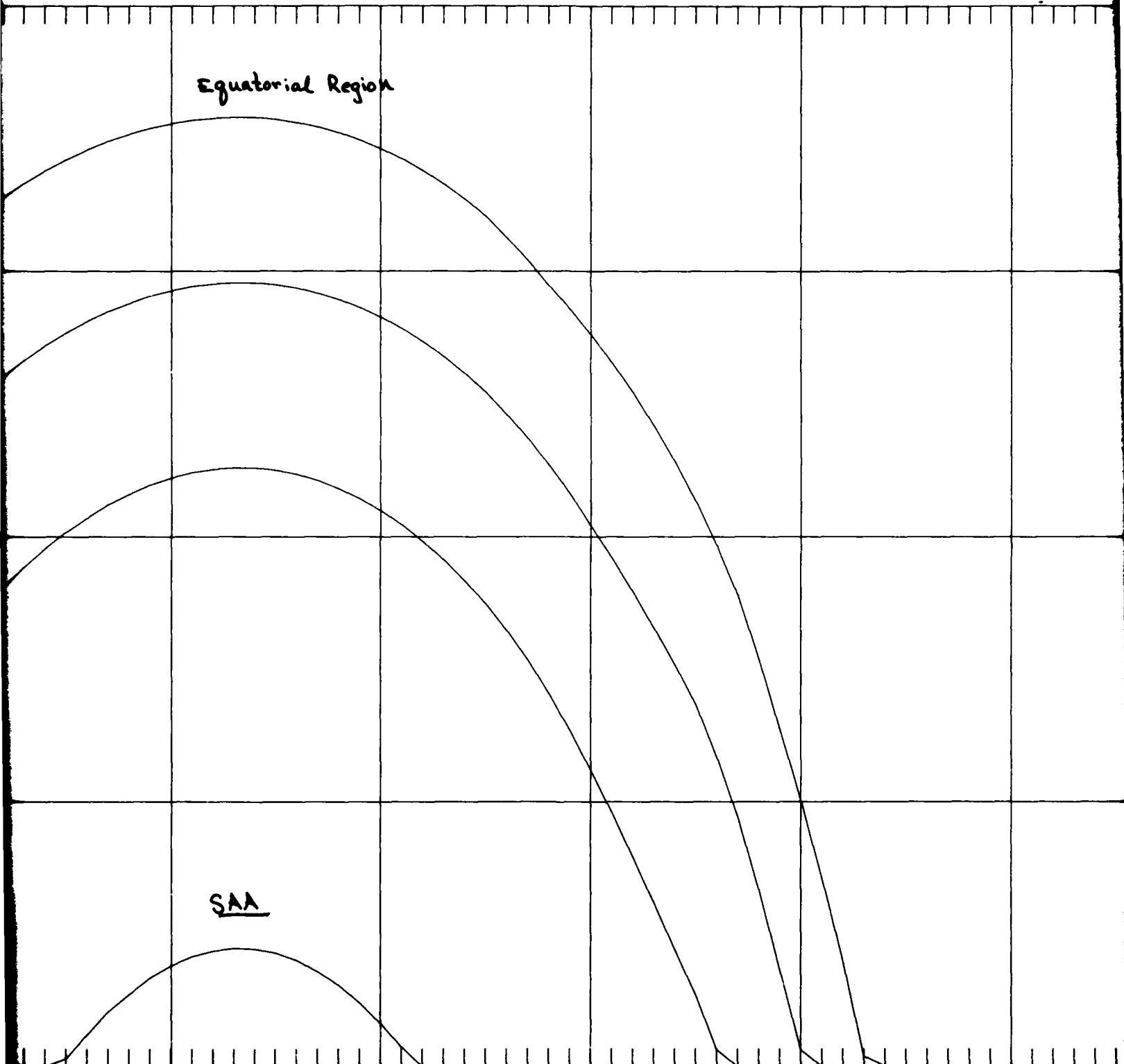
170.0

180.0

190.0

200.0

210.0



6

ORBIT: NAVELEX 5  
60 DGR/6389-6889 KM

Figure III

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 8.933318

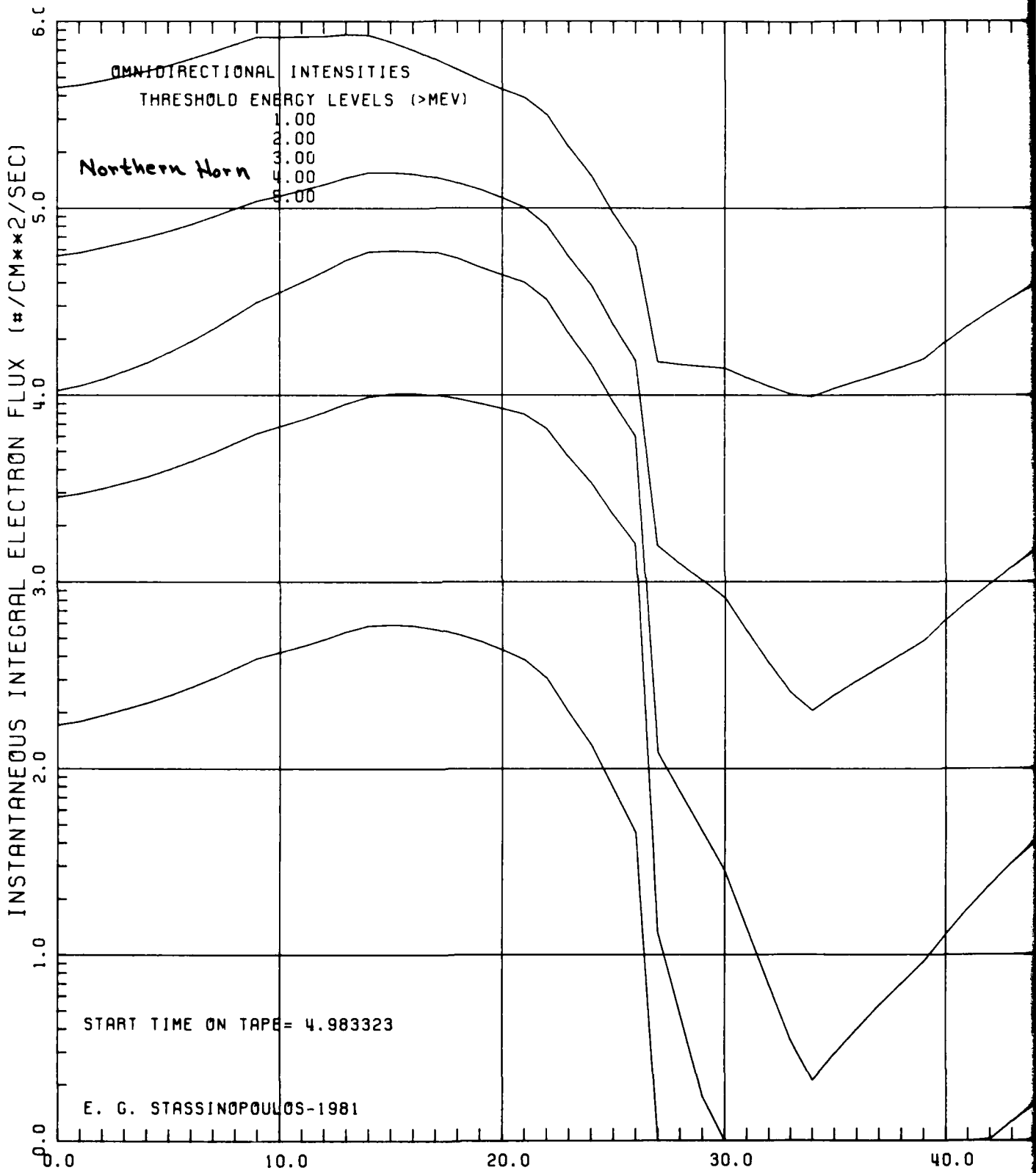
NASA-GSFC

210.0

220.0

230.0

240.0



2

Equatorial Region

SAA

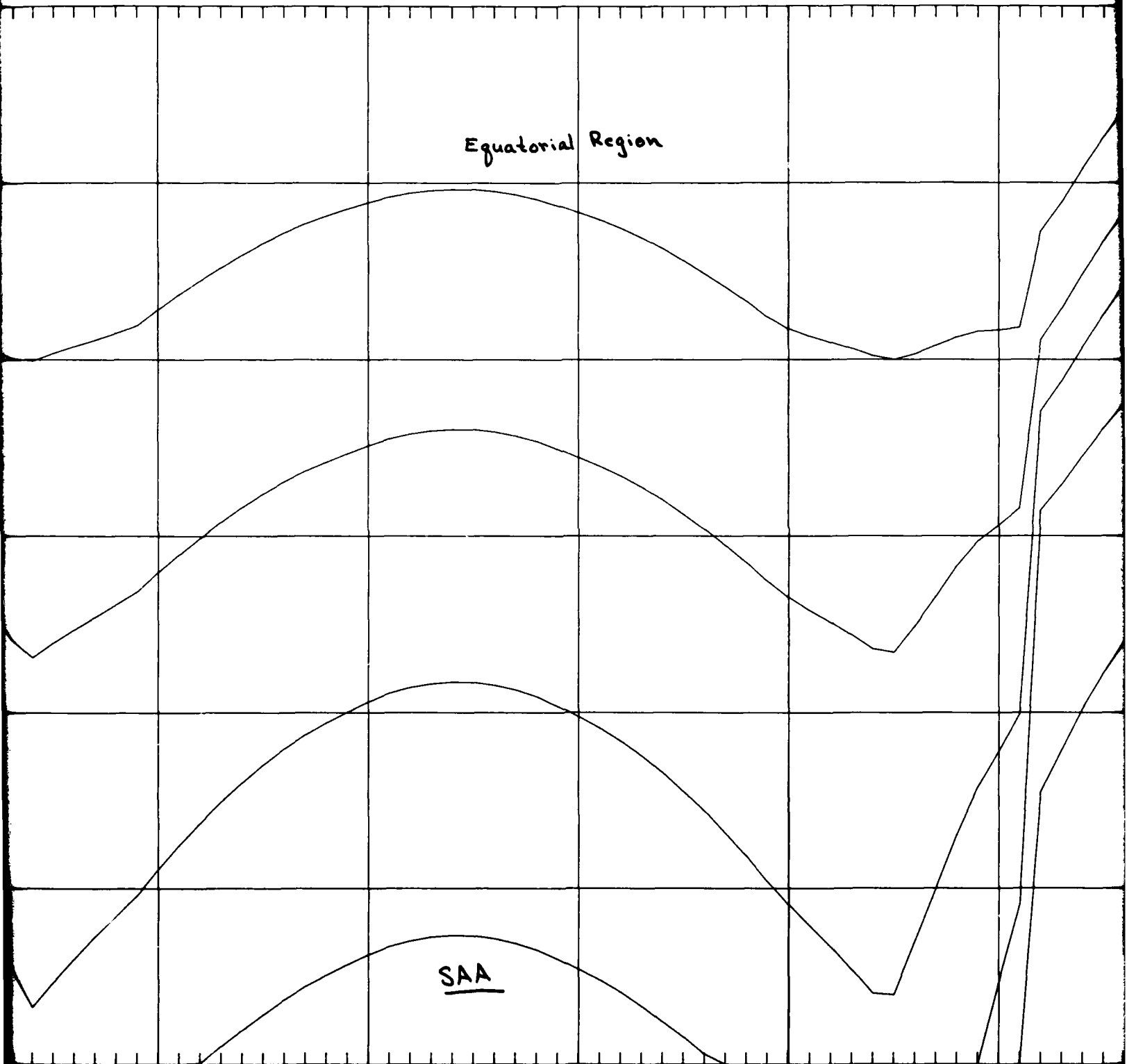
40.0

50.0

60.0

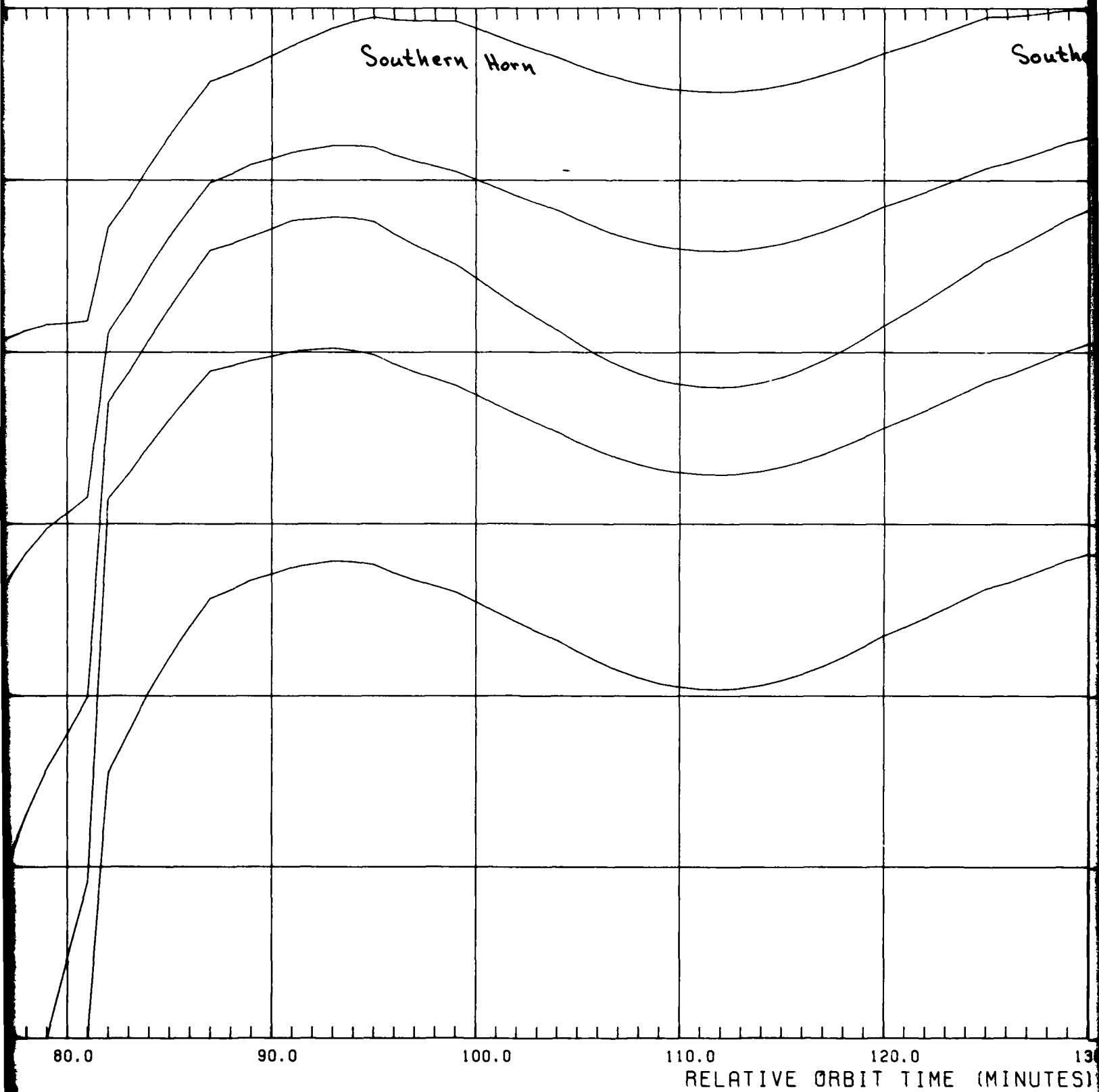
70.0

80.0



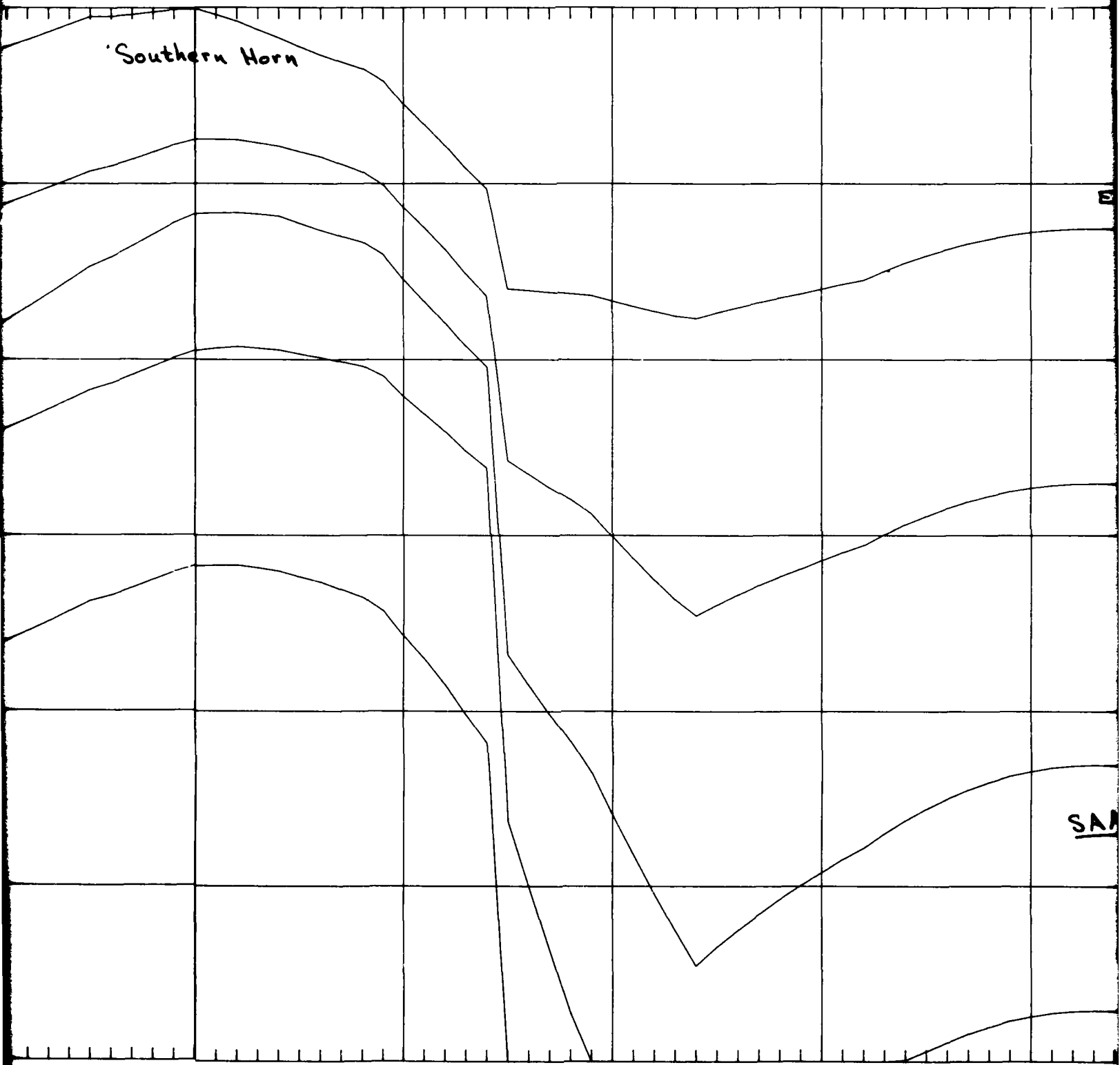


3



4

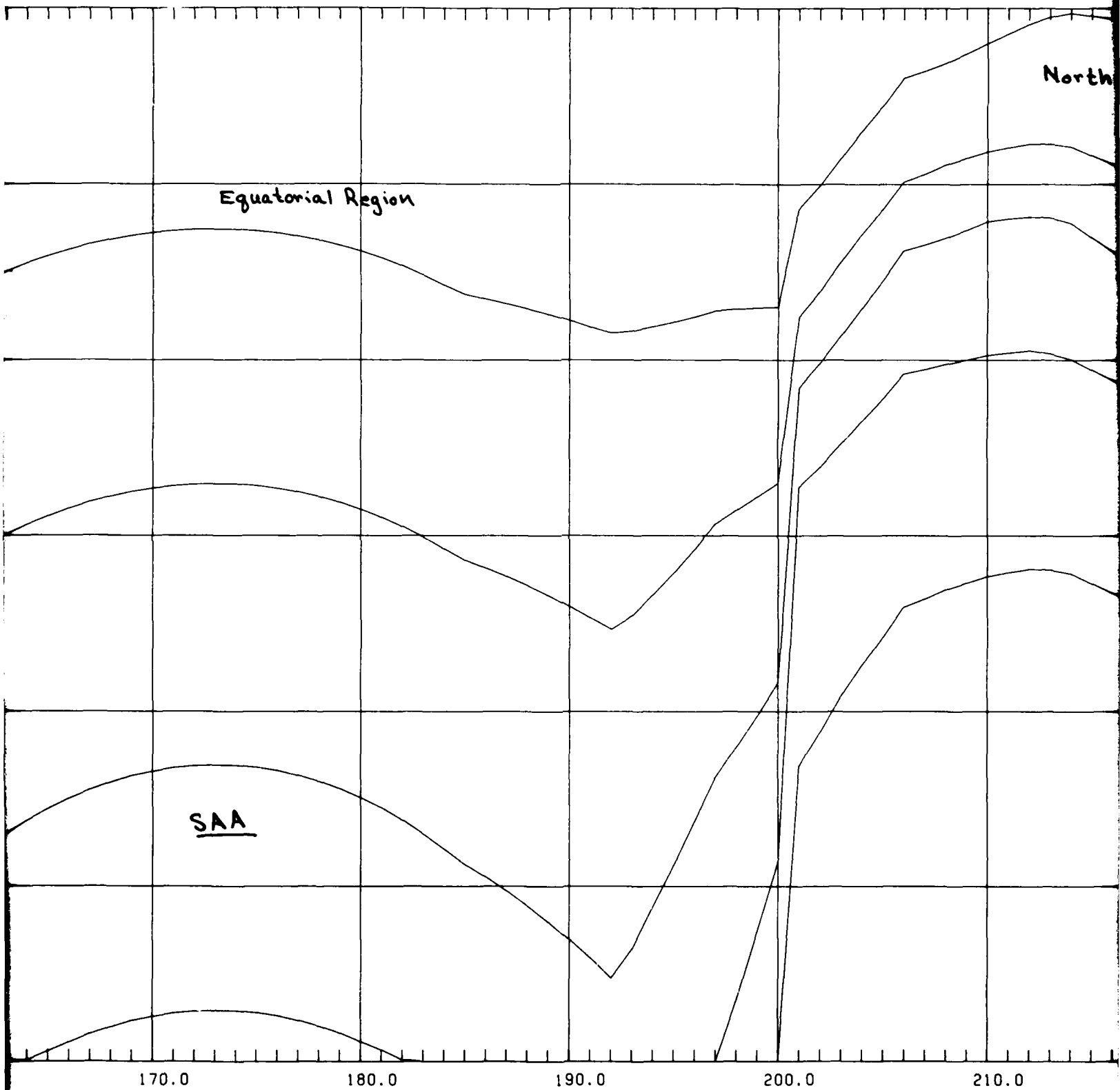
Southern Horn



SAM

0 130.0 140.0 150.0 160.0 170.0  
TIME (MINUTES)

5



6

Northern Horn

Figure 112

ORBIT: NAVELEX 5  
60 DGR/6389-6889 KM

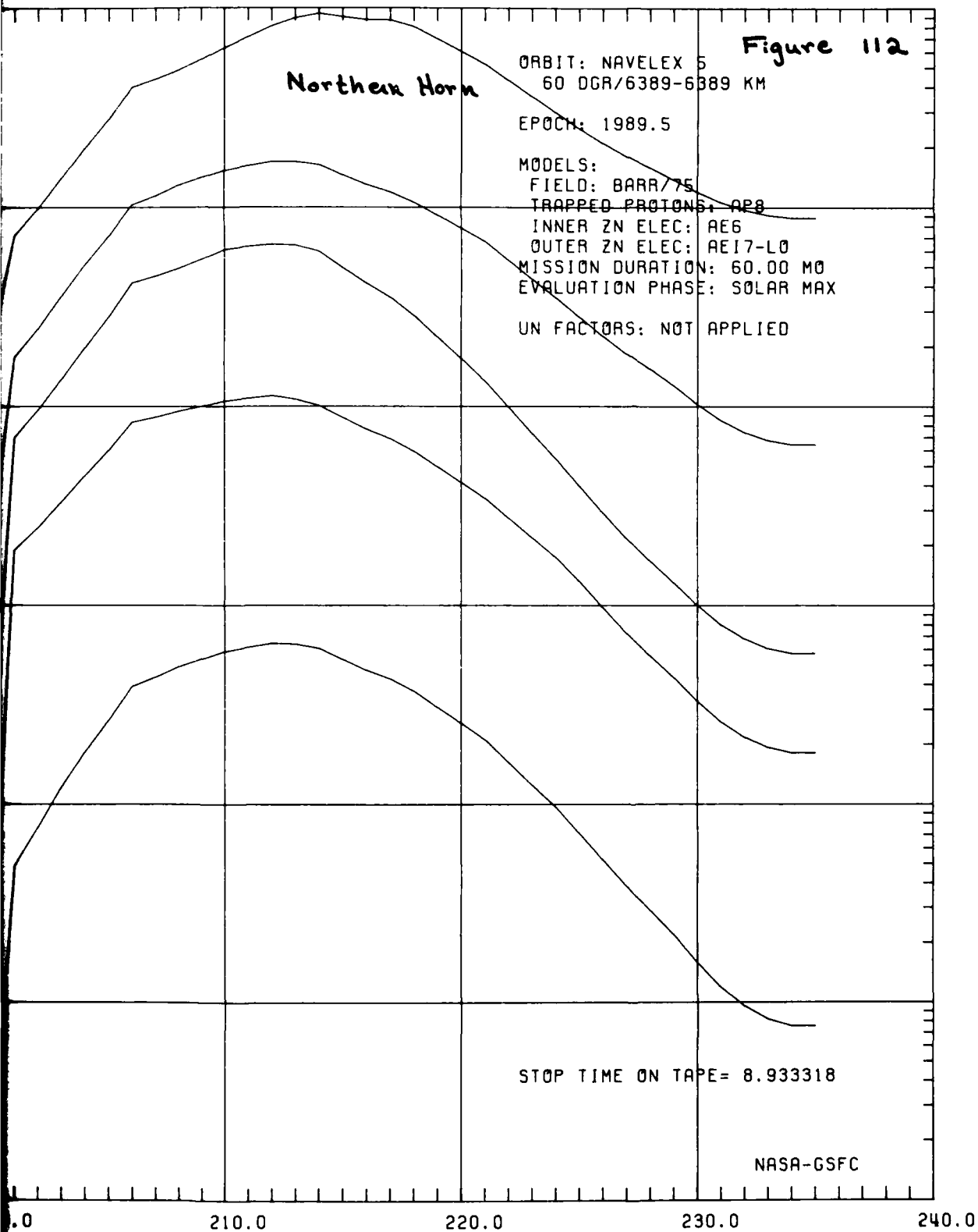
EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE= 8.933318

NASA-GSFC



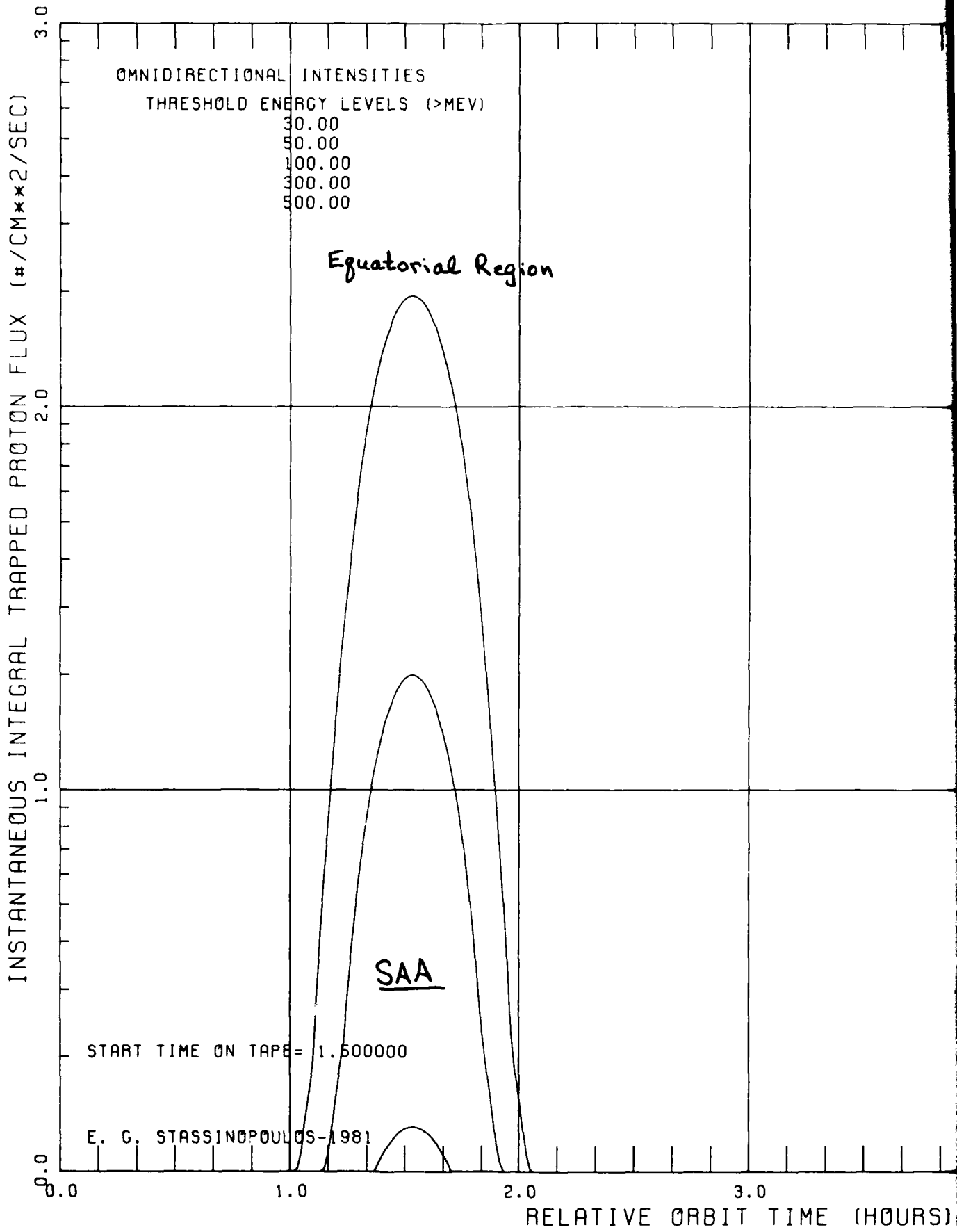


Figure 113

ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

ORBIT TIME (HOURS)

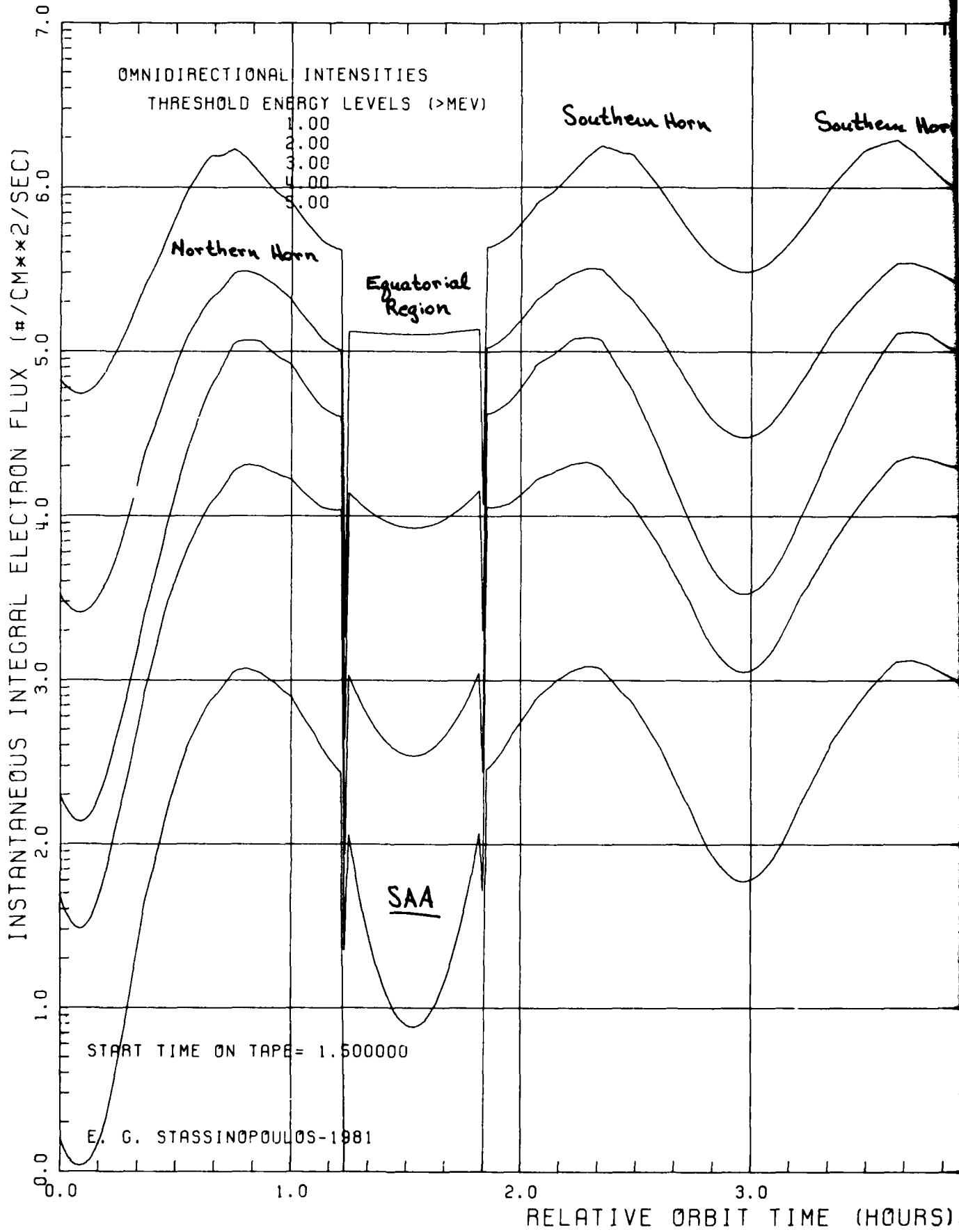


Figure 114

ORBIT: NAVELEX 6  
60 DGA/10371-10371 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Southern Horn

Northern Horn

Equatorial  
Region

SAA

STOP TIME ON TAPE = 7.483318

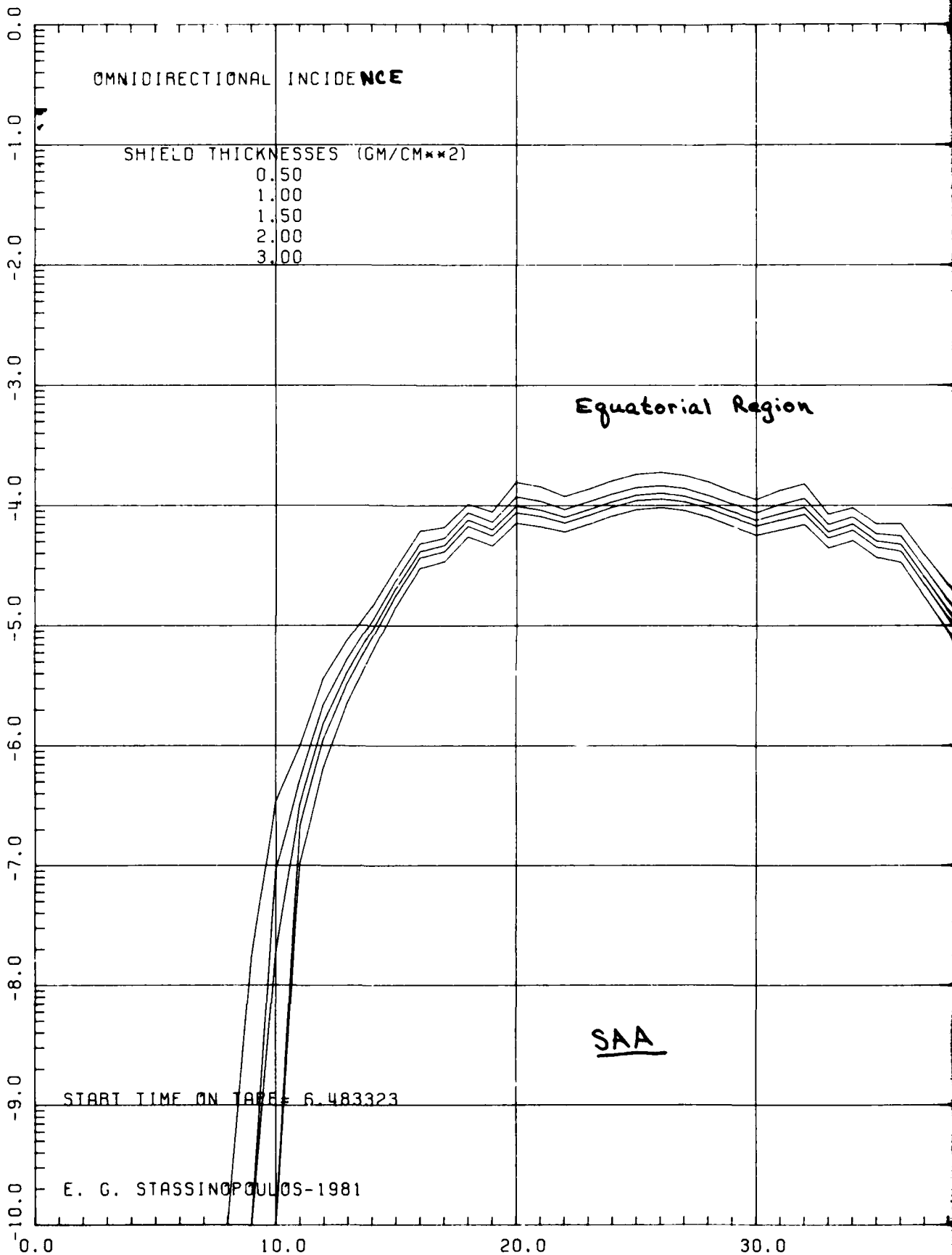
NASA-GSFC

ORBIT TIME (HOURS)



INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)

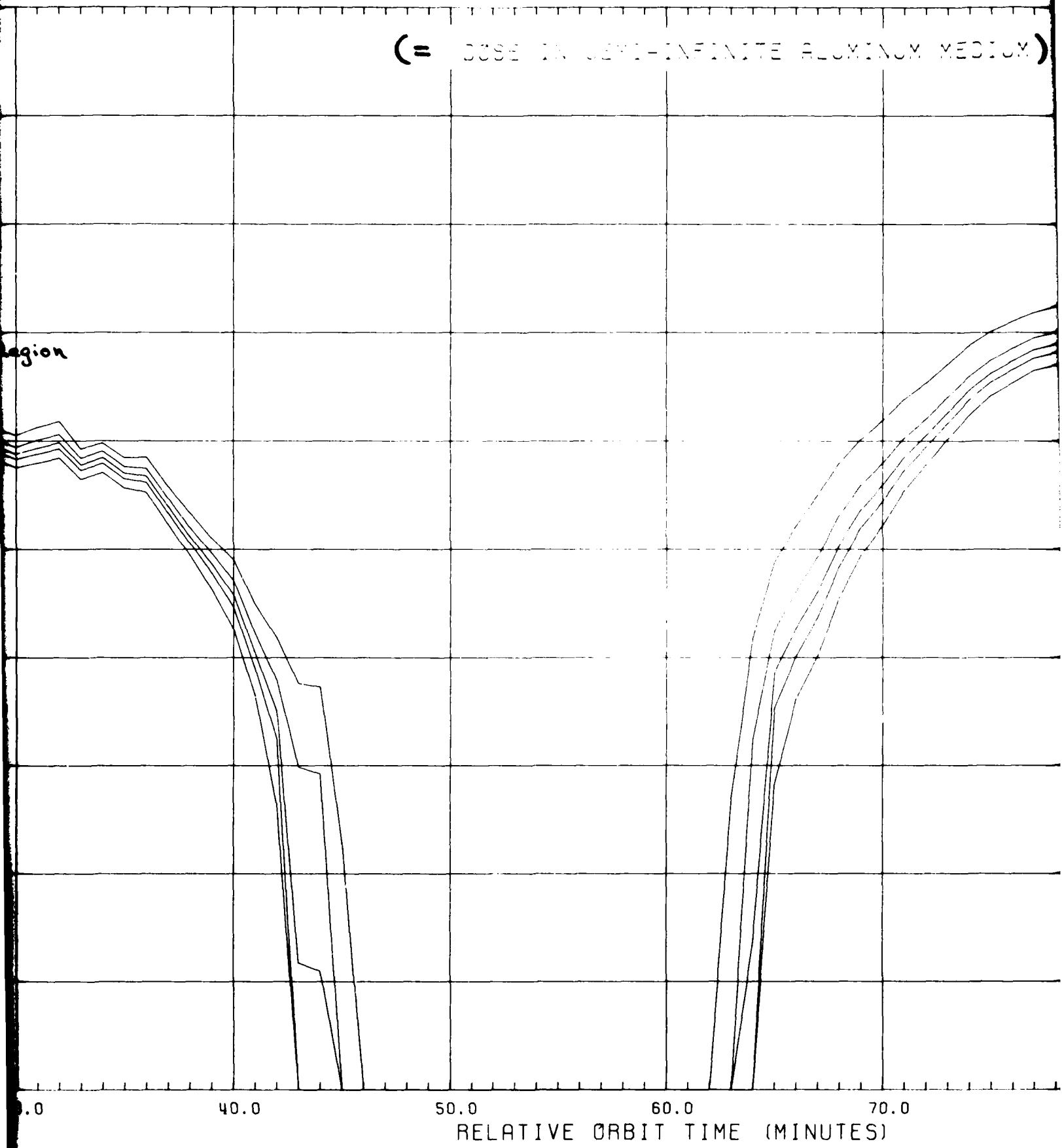


2

# DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB

(= DOSE IN SEMI-INFINITE ALUMINUM MEDIUM)

Region



OF FINITE ALUMINUM SLAB SHIELDS

FINITE ALUMINUM MEDIUM)

ORBIT: NAVELEX 1  
60 DGR/1667-15

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS  
INNER ZN ELEC:  
OUTER ZN ELEC:

MISSION DURATION:  
EVALUATION PHASE:

UN FACTORS: NOT A

Equatorial Region

SAA

STOP TIME ON TAPE

70.0

80.0

90.0

100.0

110.0

ME (MINUTES)

Figure 115

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

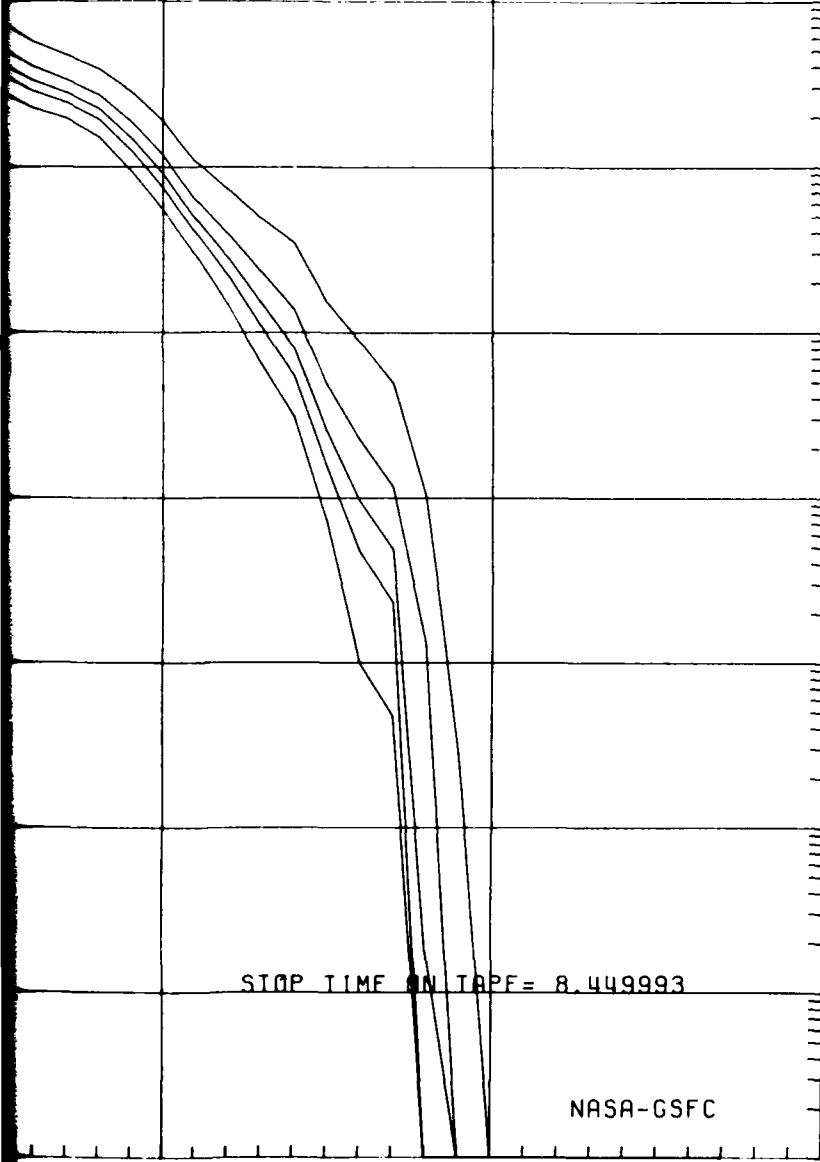
INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED



STOP TIME ON TAPE = 8.449993

NASA-GSFC

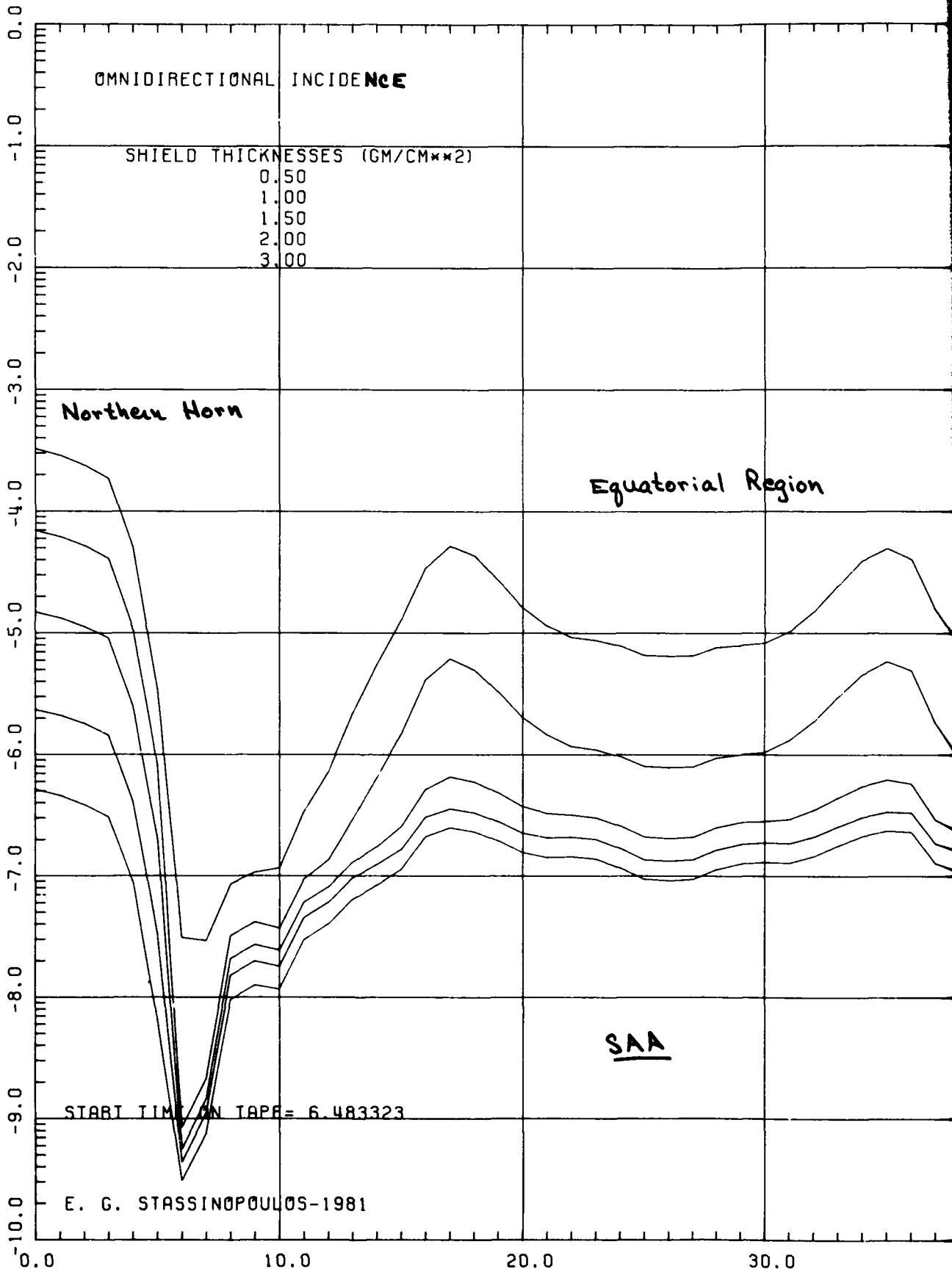
100.0

110.0

120.0

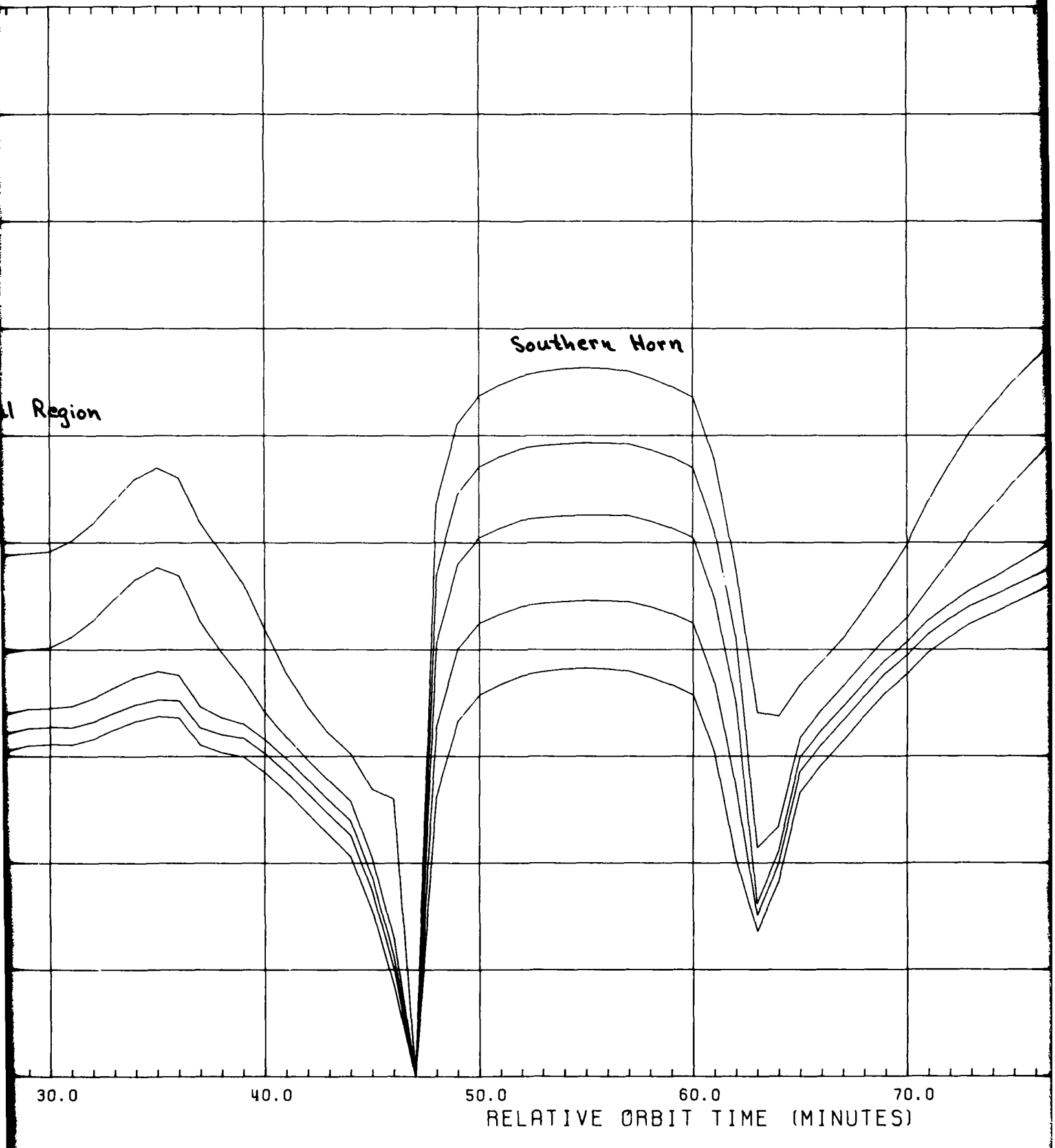
INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



1  
2

# DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM S



31

# FINITE ALUMINUM SLAB SHIELDS

Figure

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L  
MISSION DURATION: 60.0  
EVALUATION PHASE: SOLAR  
UN FACTORS: NOT APPLIED

Equatorial Region

No

SAA

STOP TIME ON TAPE = 8.4

NA

MINUTES) 70.0 80.0 90.0 100.0 110.0

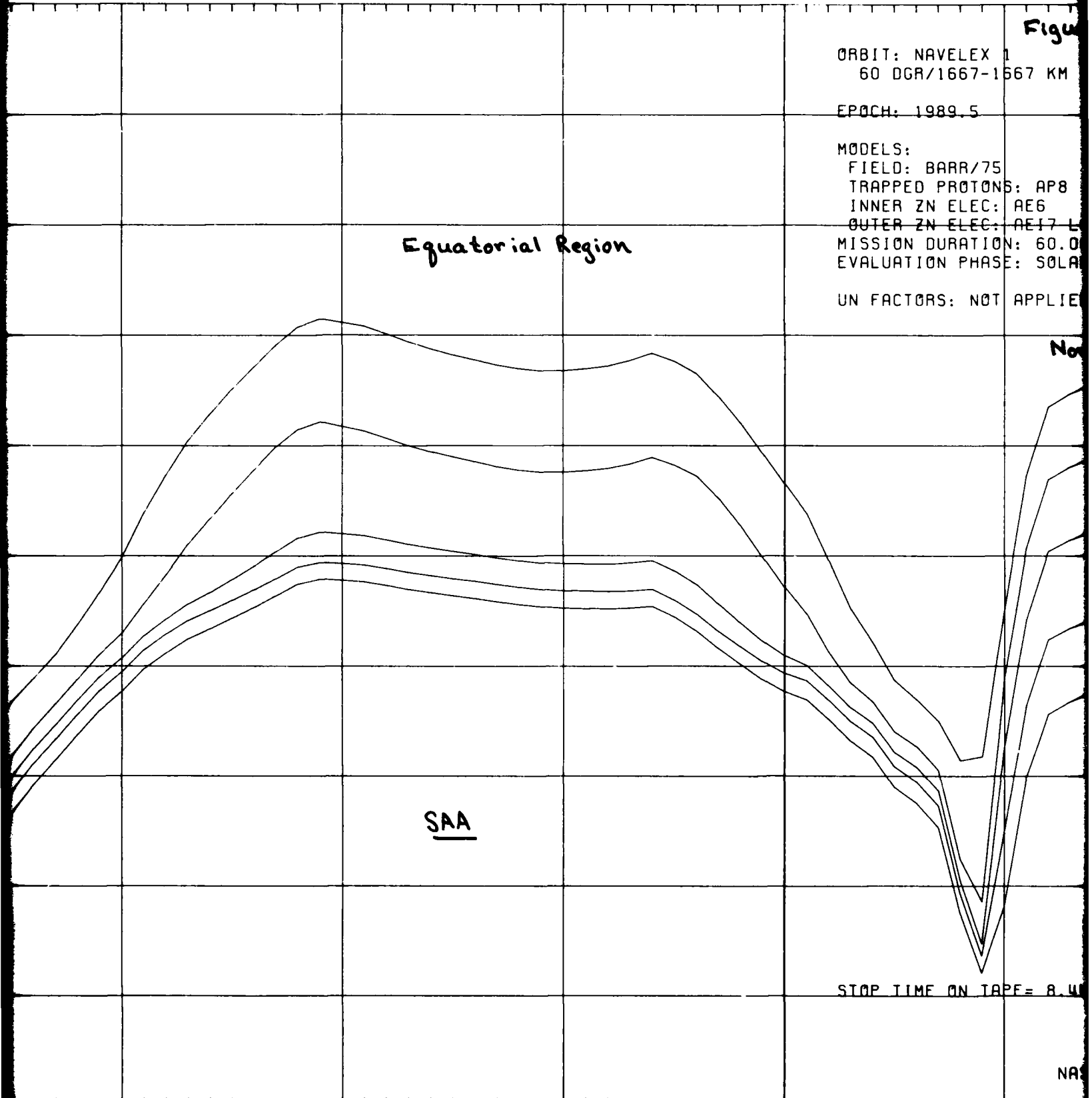


Figure 116

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

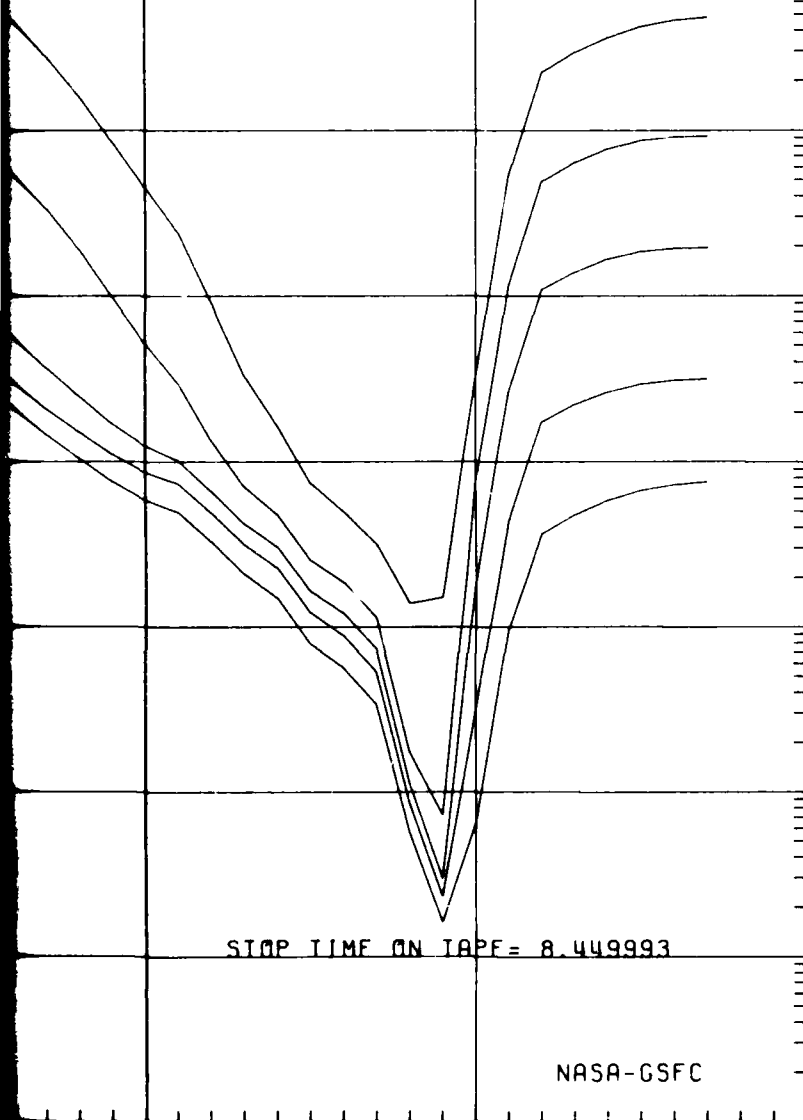
EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn



STOP TIME ON TAPE = 8.449993

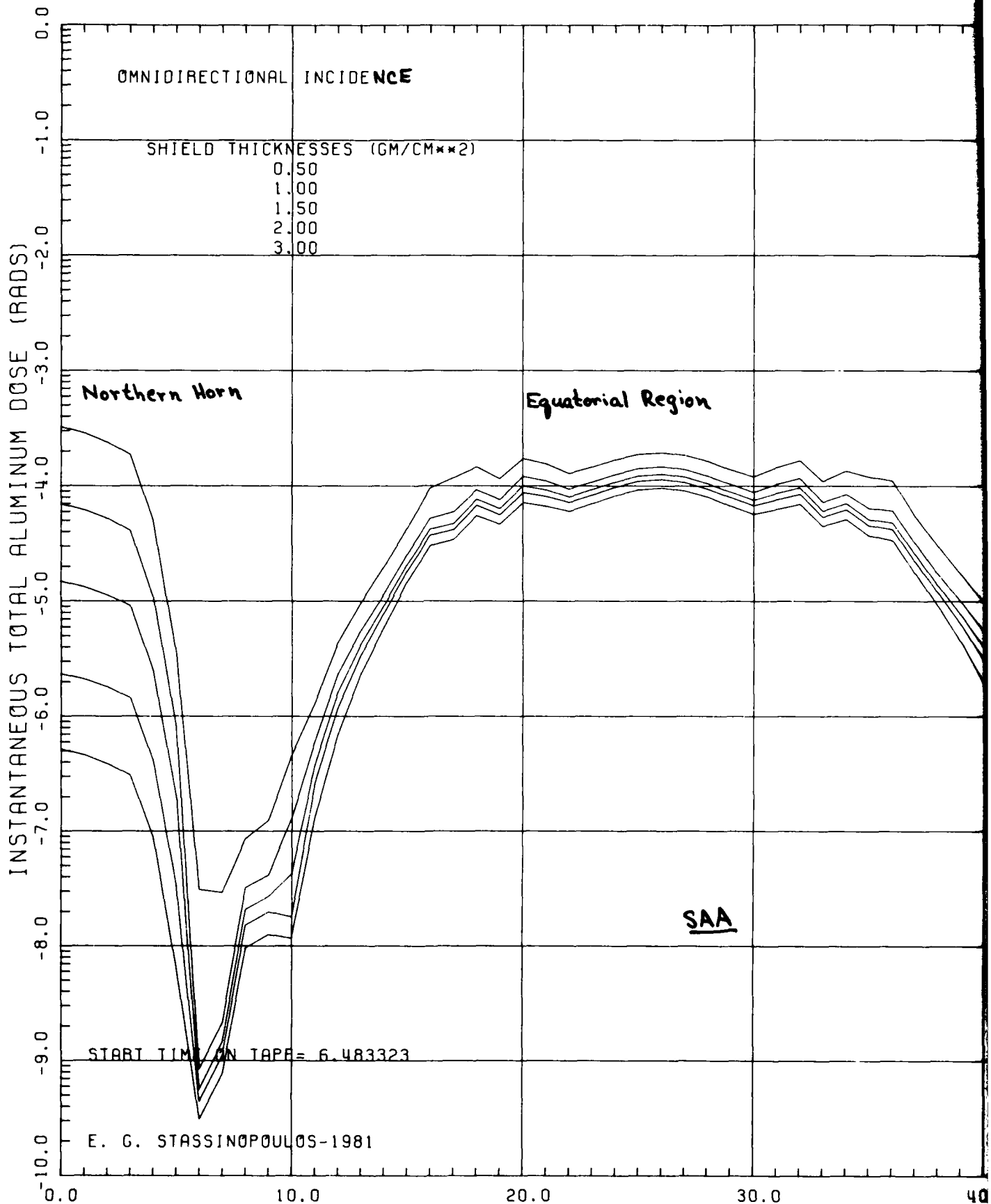
NASA-GSFC

100.0

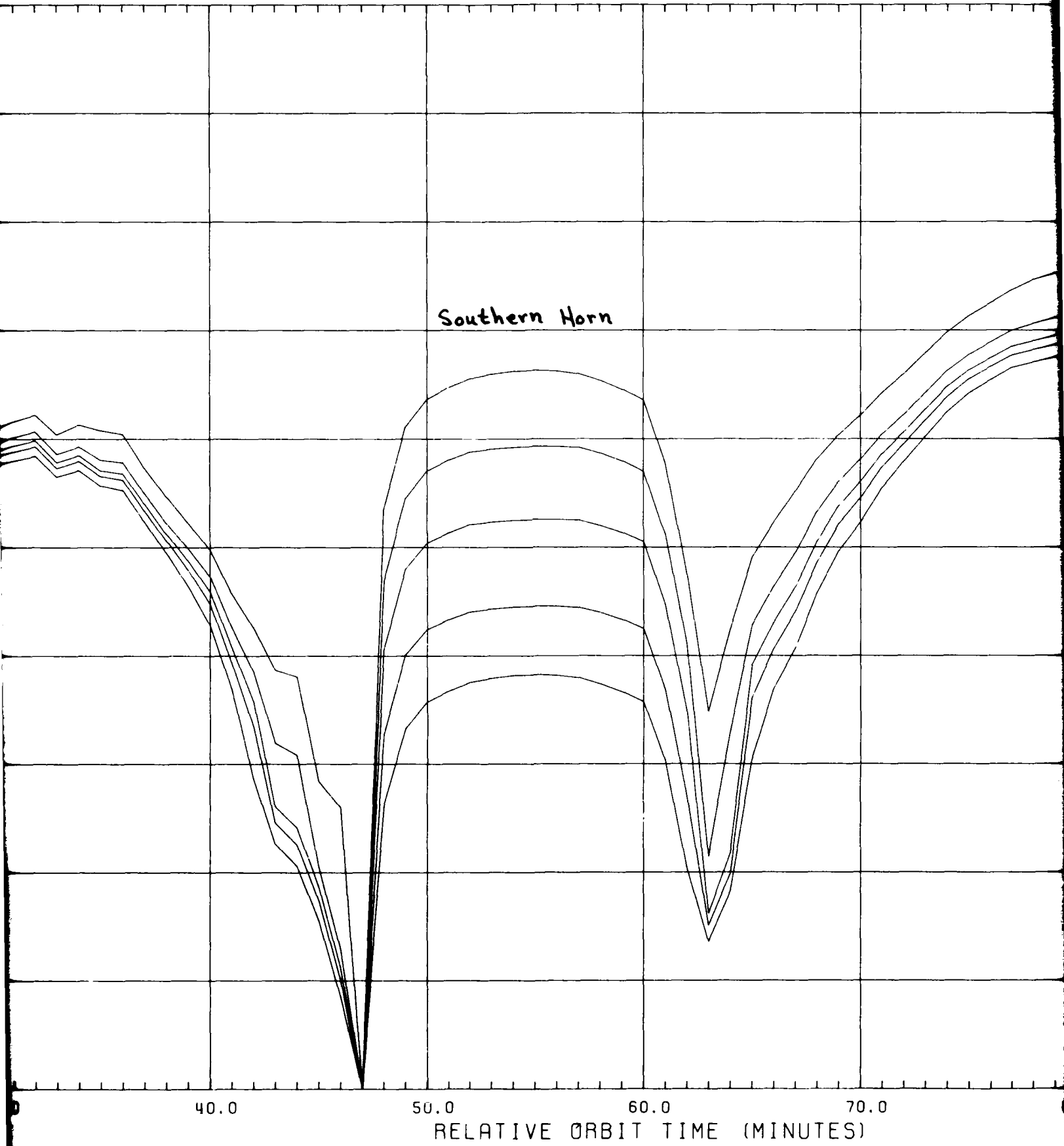
110.0

120.0





DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB



# E ALUMINUM SLAB SHIELDS

Figure

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

Northern

SAA

STOP TIME ON TAPE = 8.449993

NASA-GSF

70.0 80.0 90.0 100.0 110.0  
(TES)

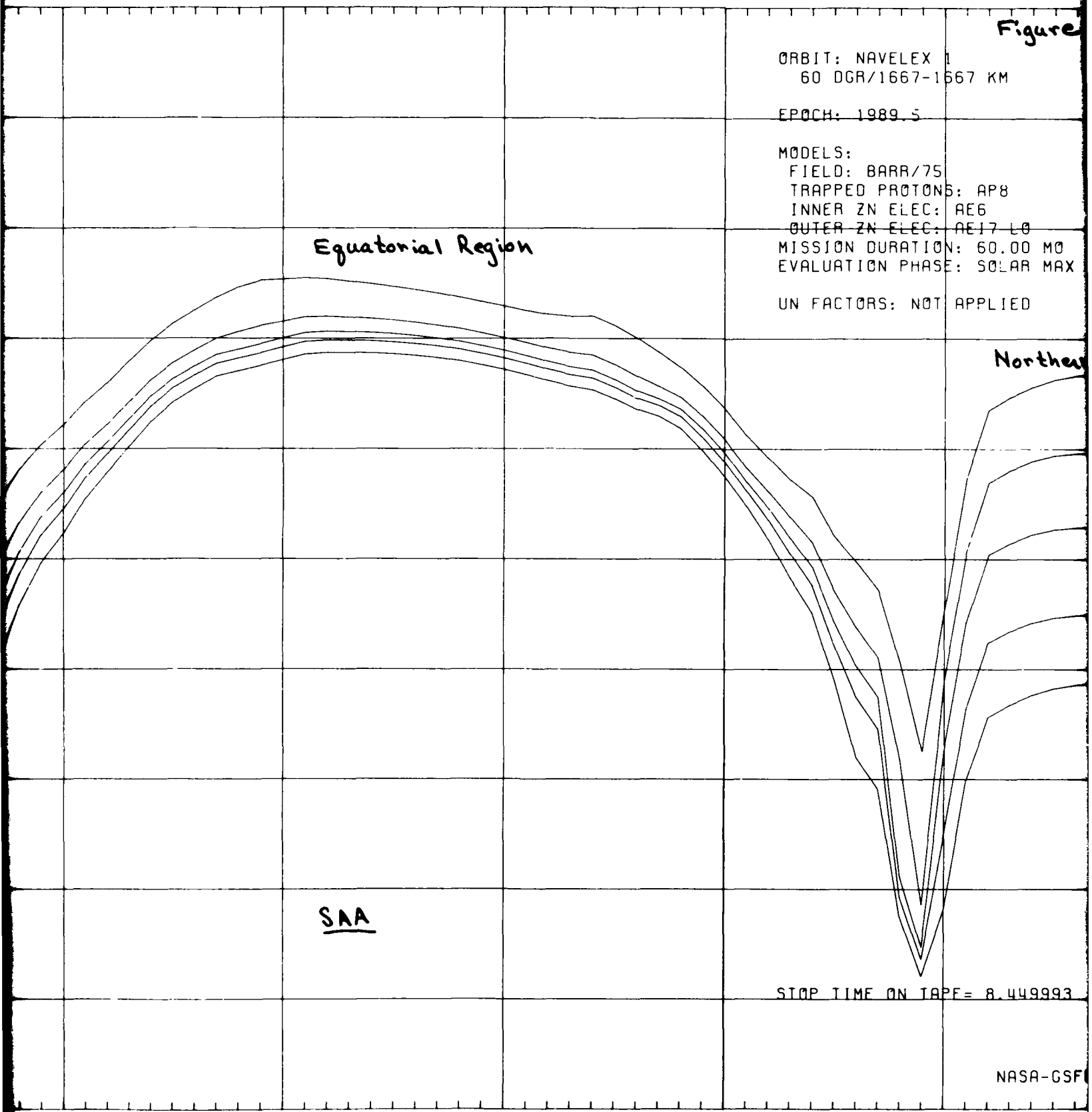


Figure 117

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

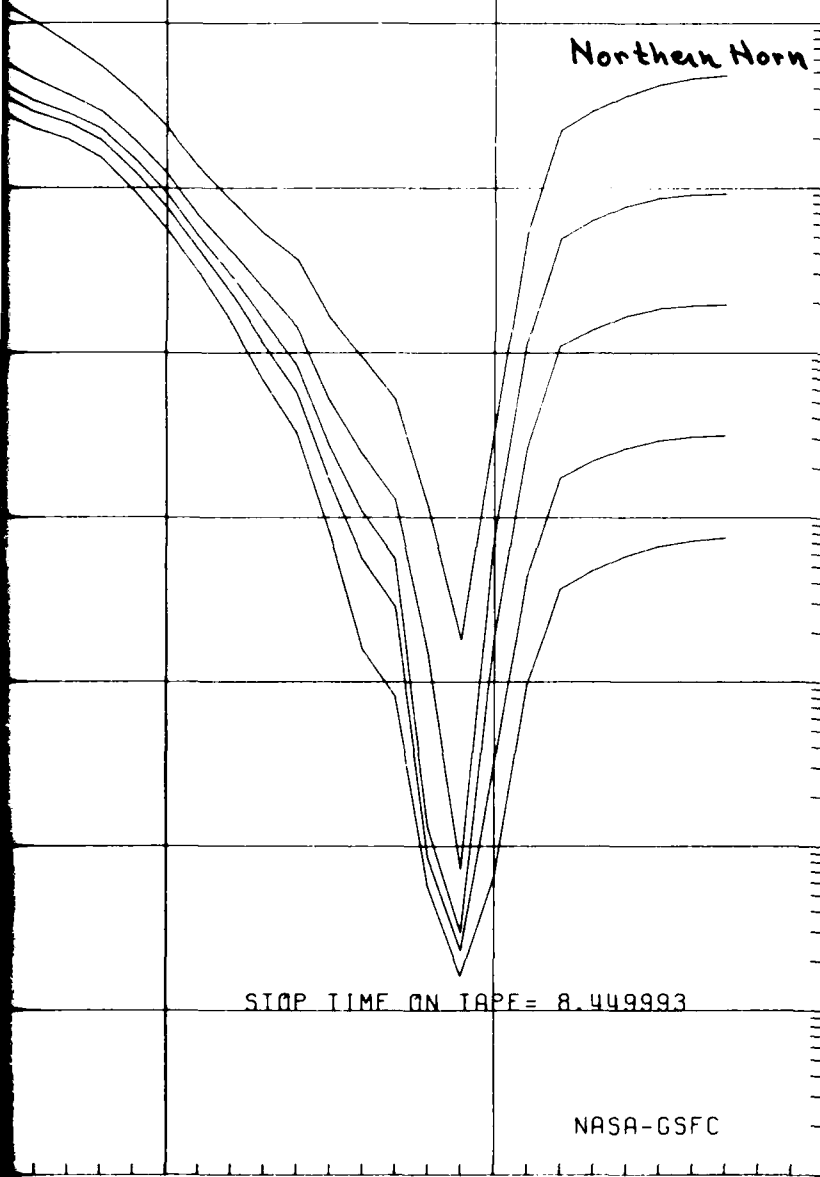
OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

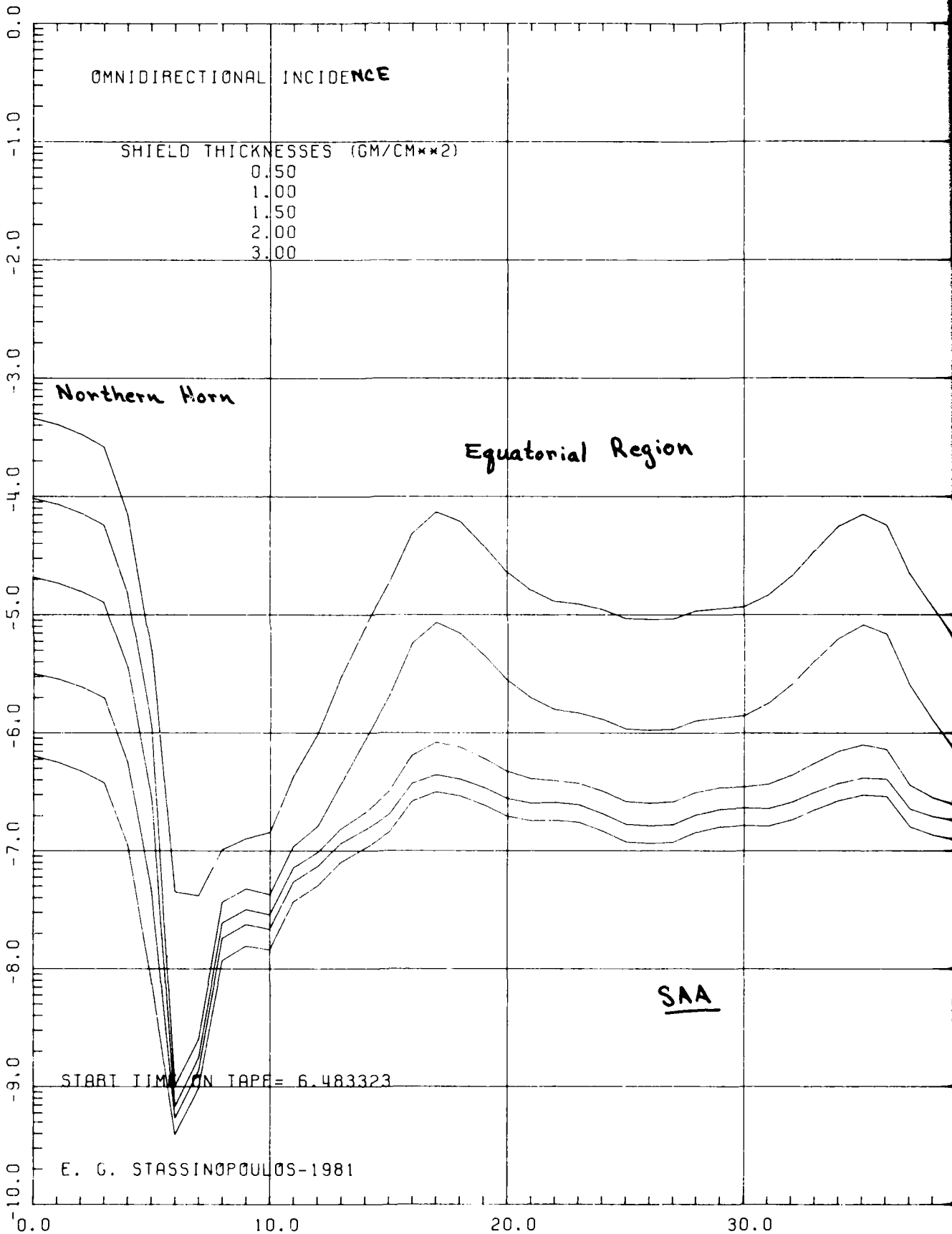


STOP TIME ON TAPE = 8.449993

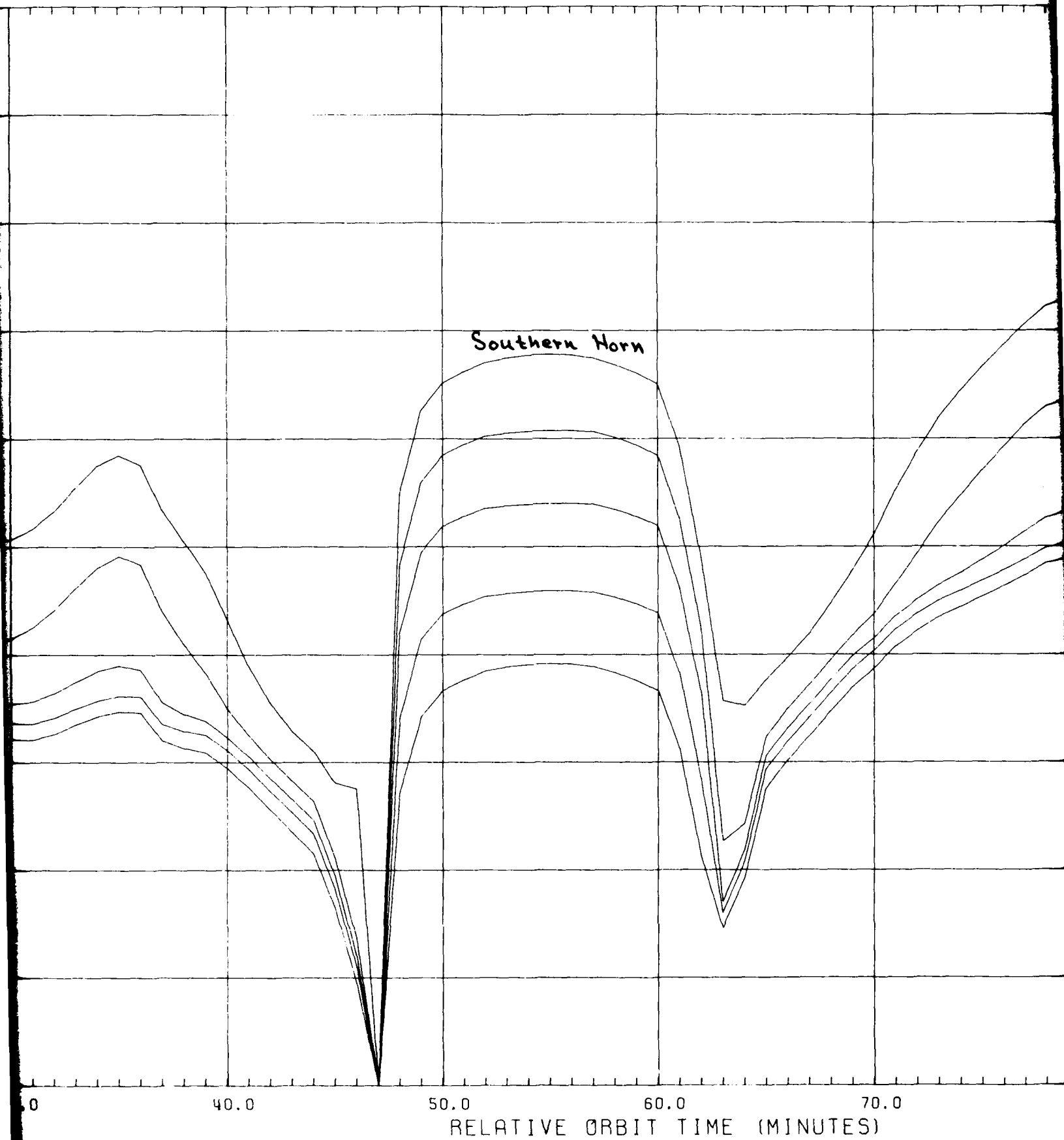
NASA-GSFC

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



IUM

Figure 118

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

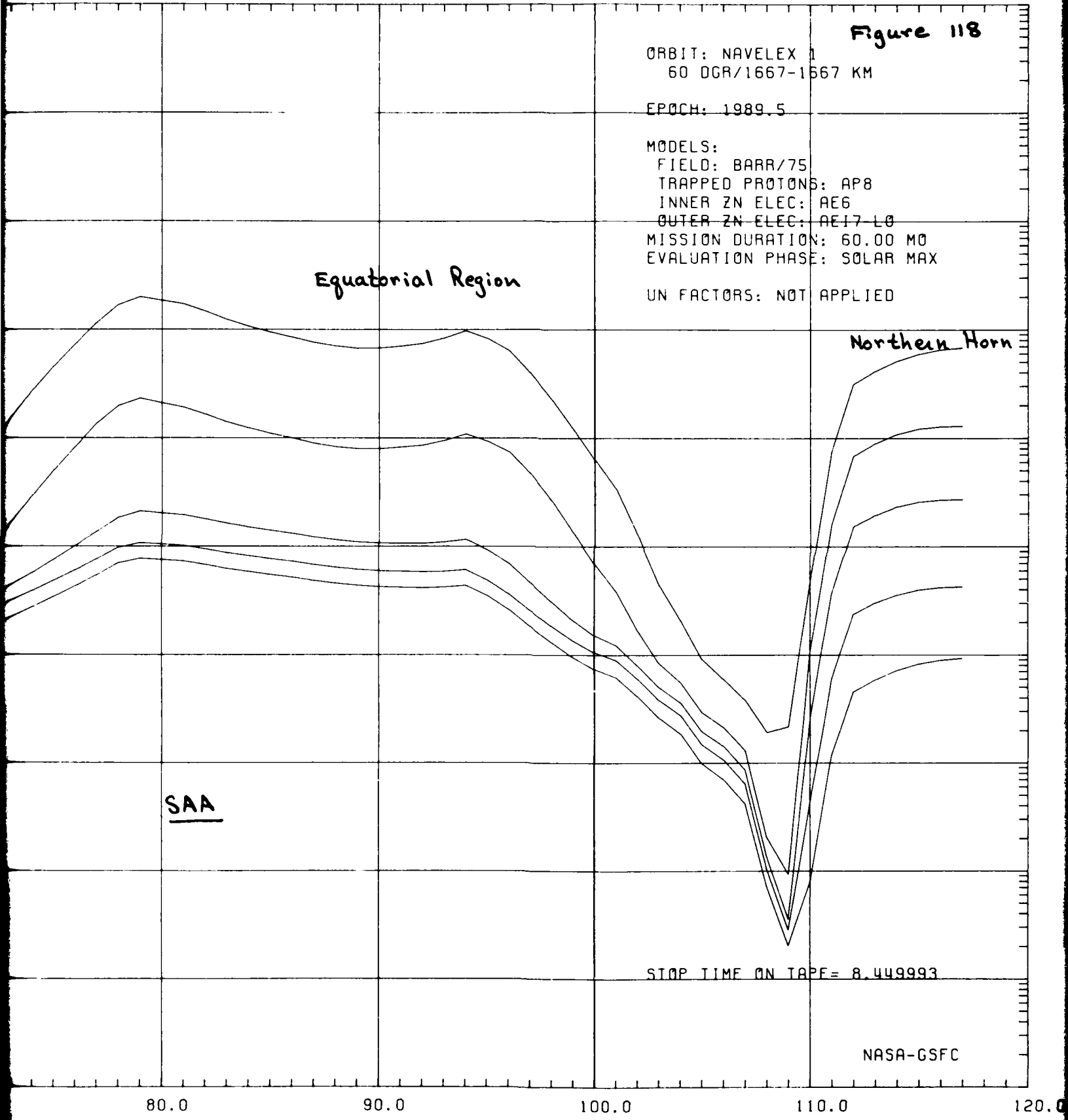
Equatorial Region

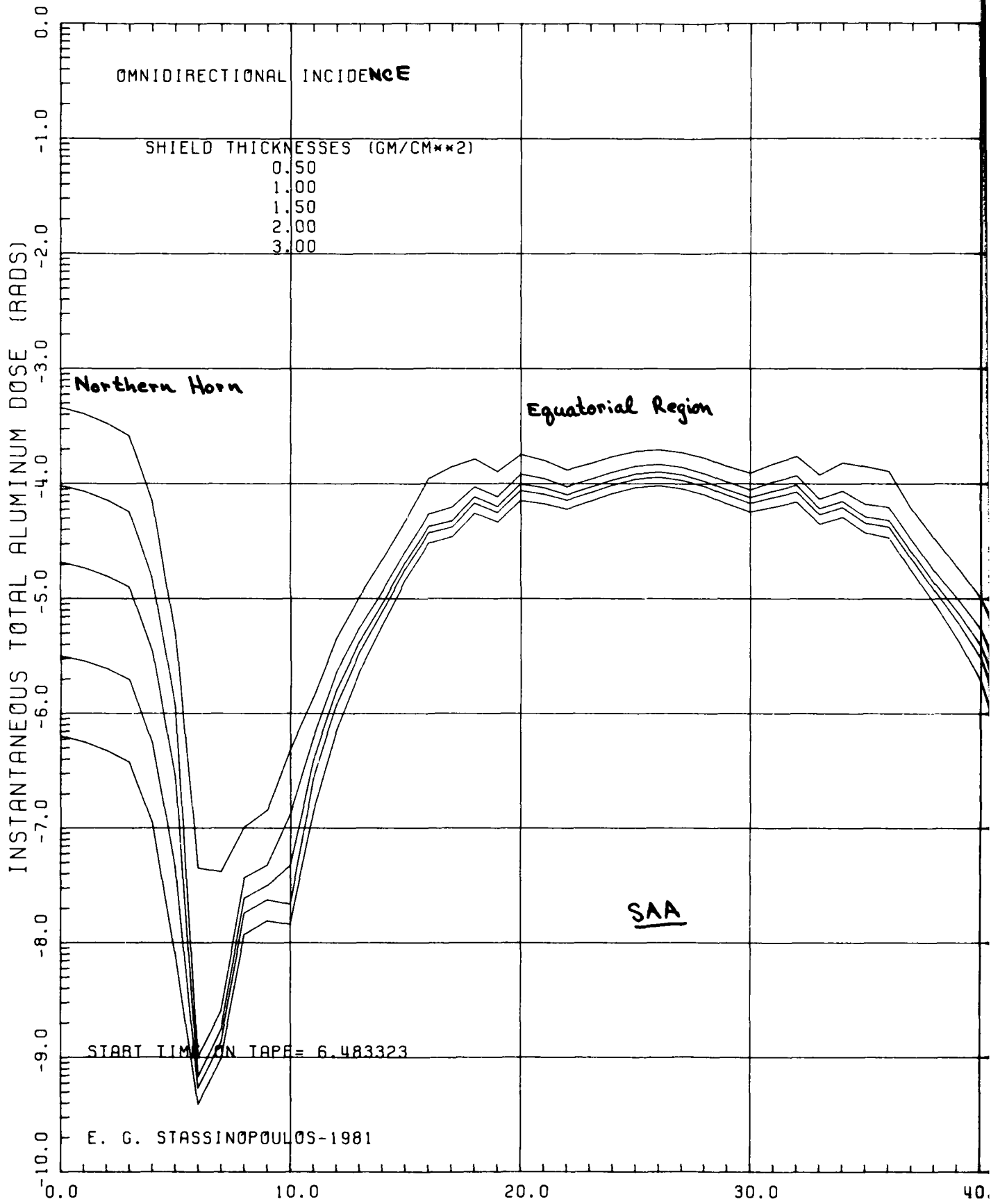
Northern Horn

SAA

STOP TIME ON TAPE= 8.449993

NASA-GSFC

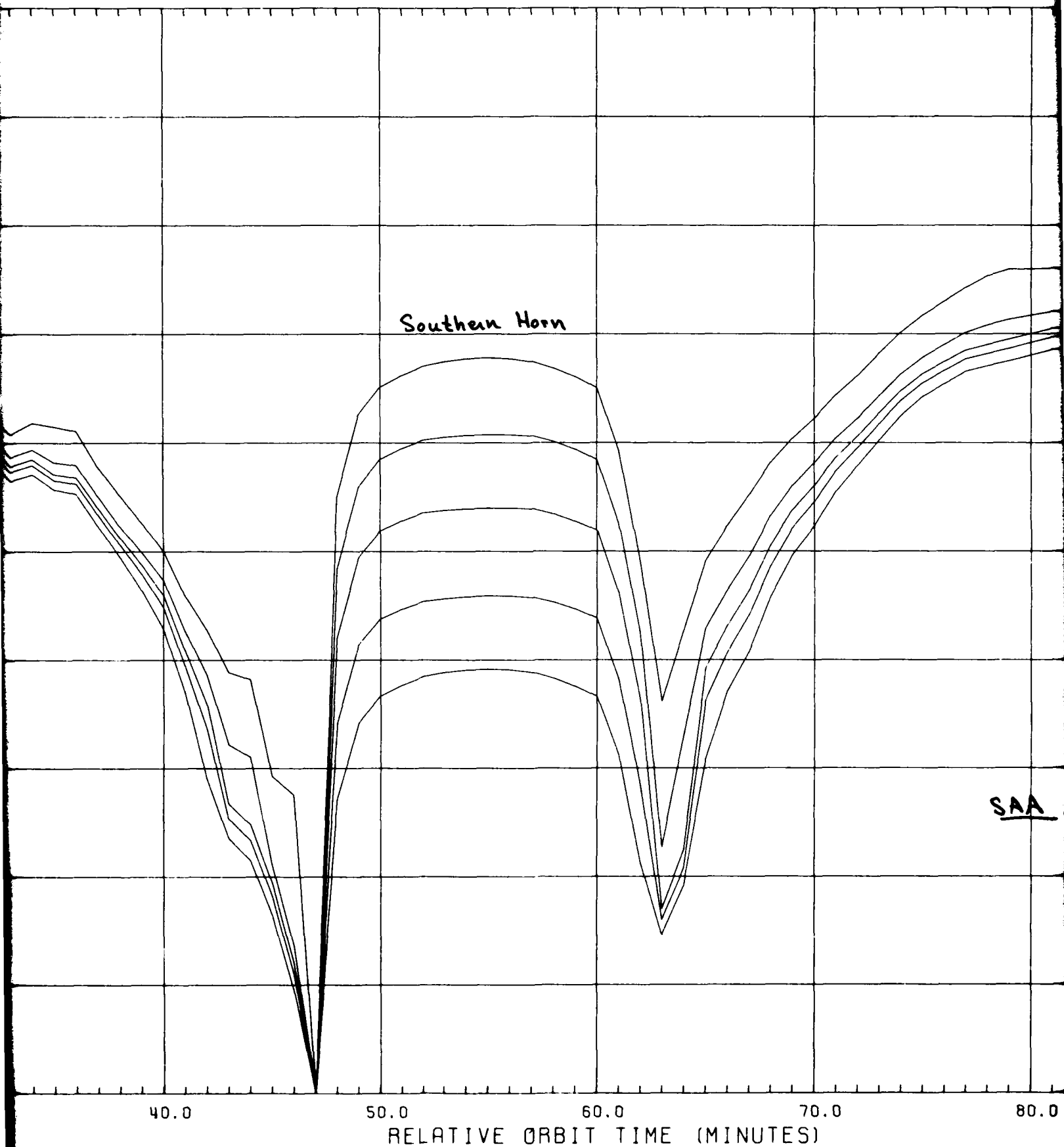






2

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



Southern Horn

SAA

UM

Figure 119

ORBIT: NAVELEX 1  
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

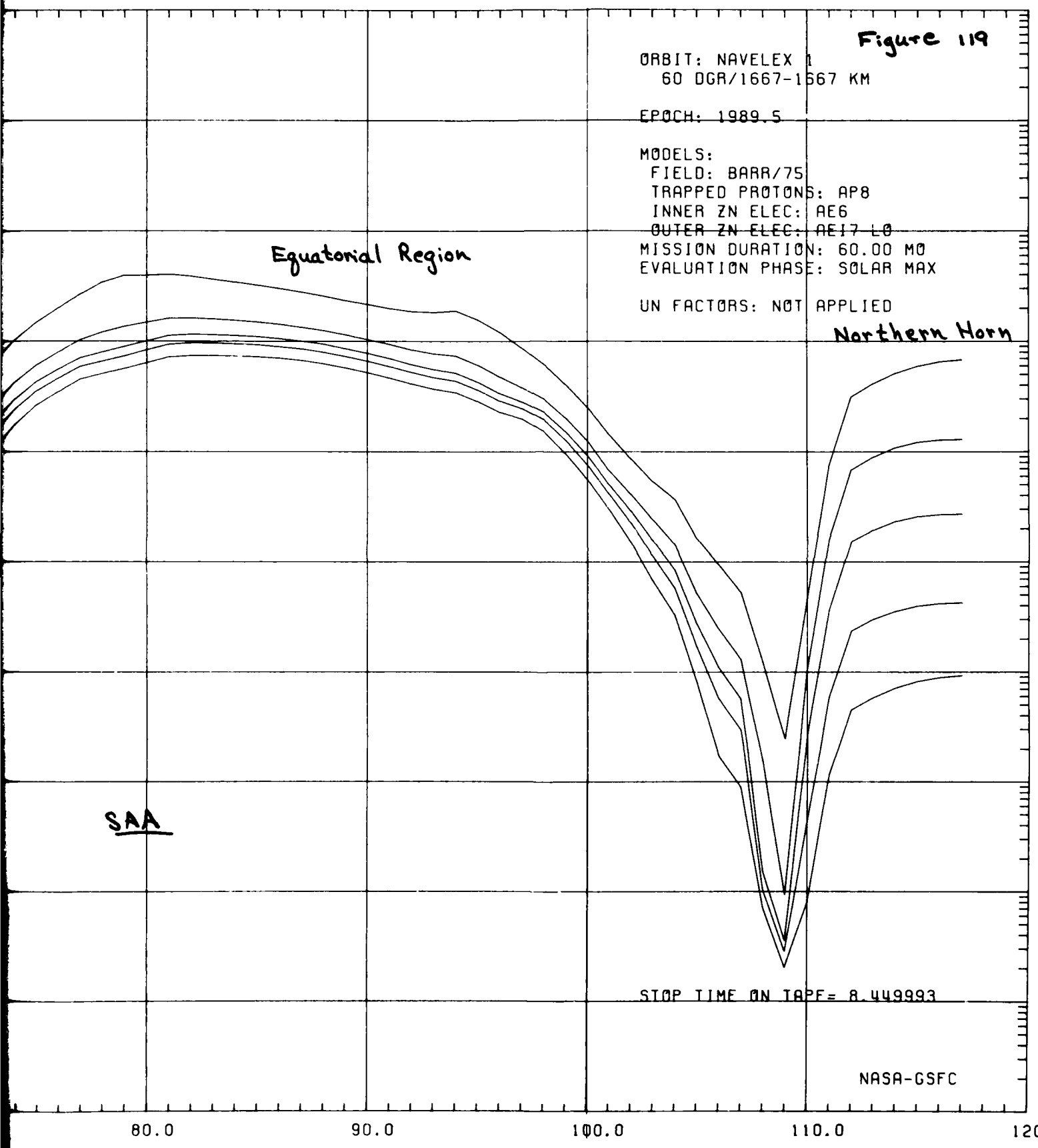
Equatorial Region

Northern Horn

SAA

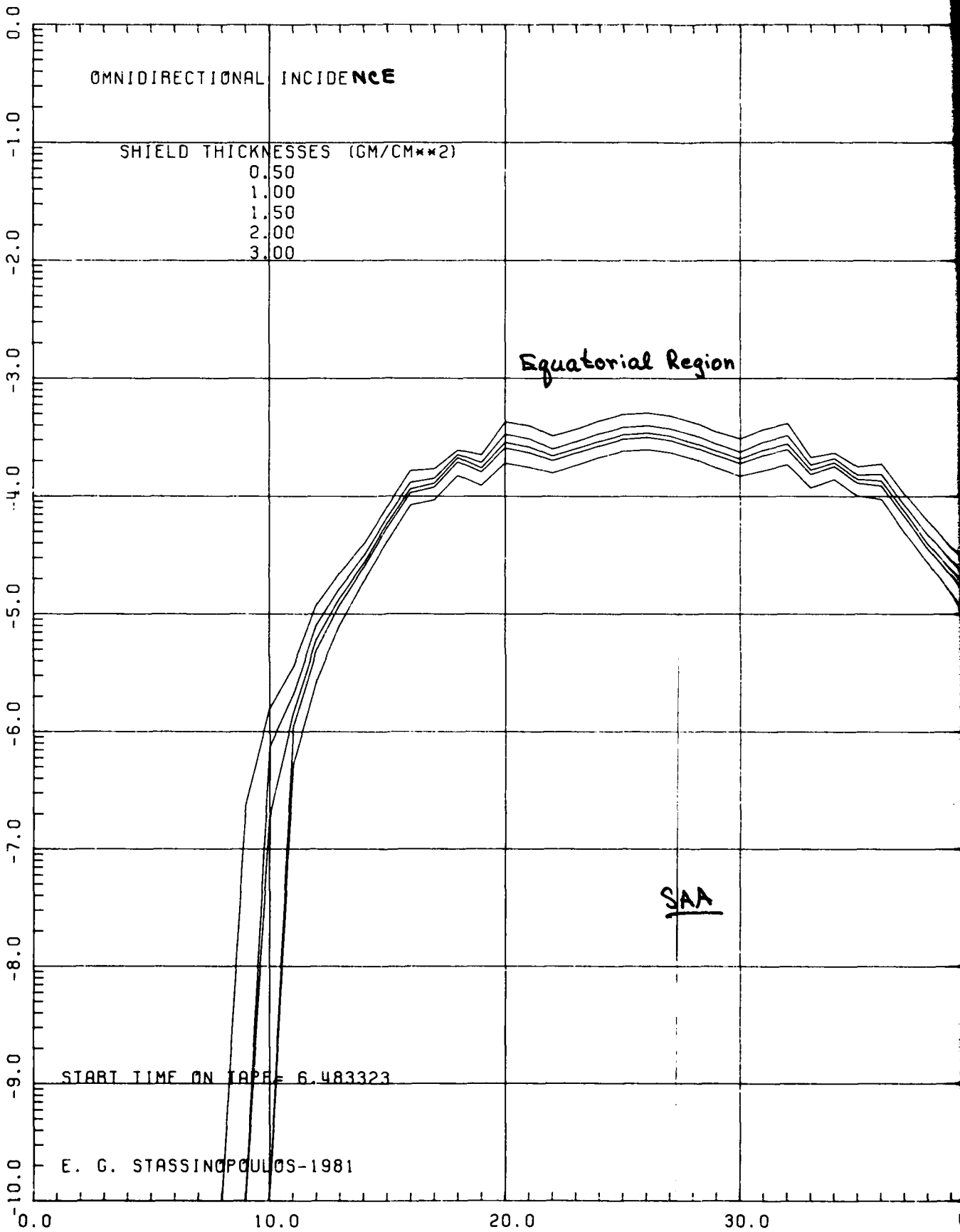
STOP TIME ON TAPE = 8.449993

NASA-GSFC



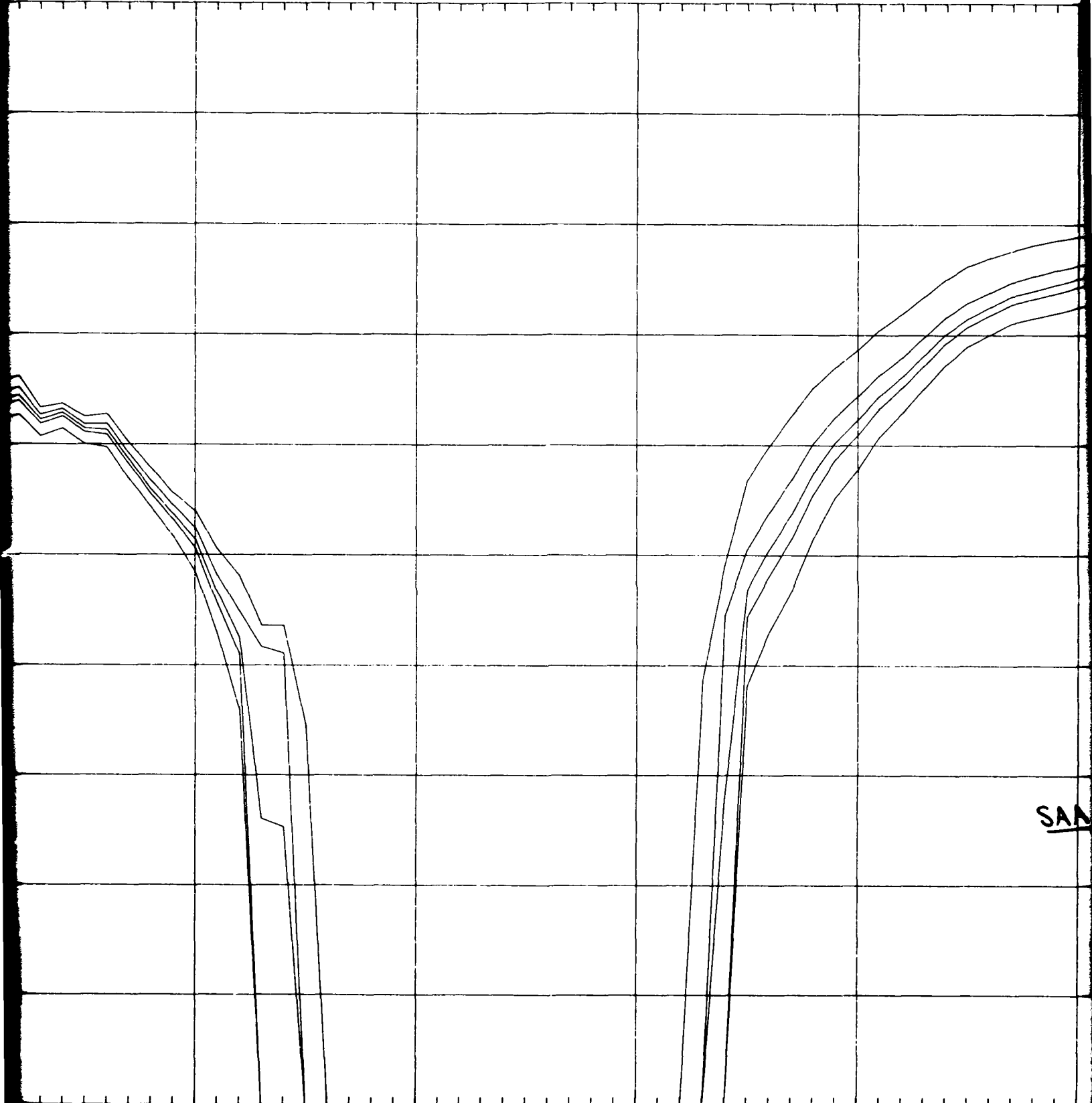
INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



'2

DOSE AT CENTER OF ALUMINUM SPHERES



40.0

50.0

60.0

70.0

80.0

RELATIVE ORBIT TIME (MINUTES)

SAN

Figure 120

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

SAA

STOP TIME ON TAPE = 8.449993

NASA-GSFC

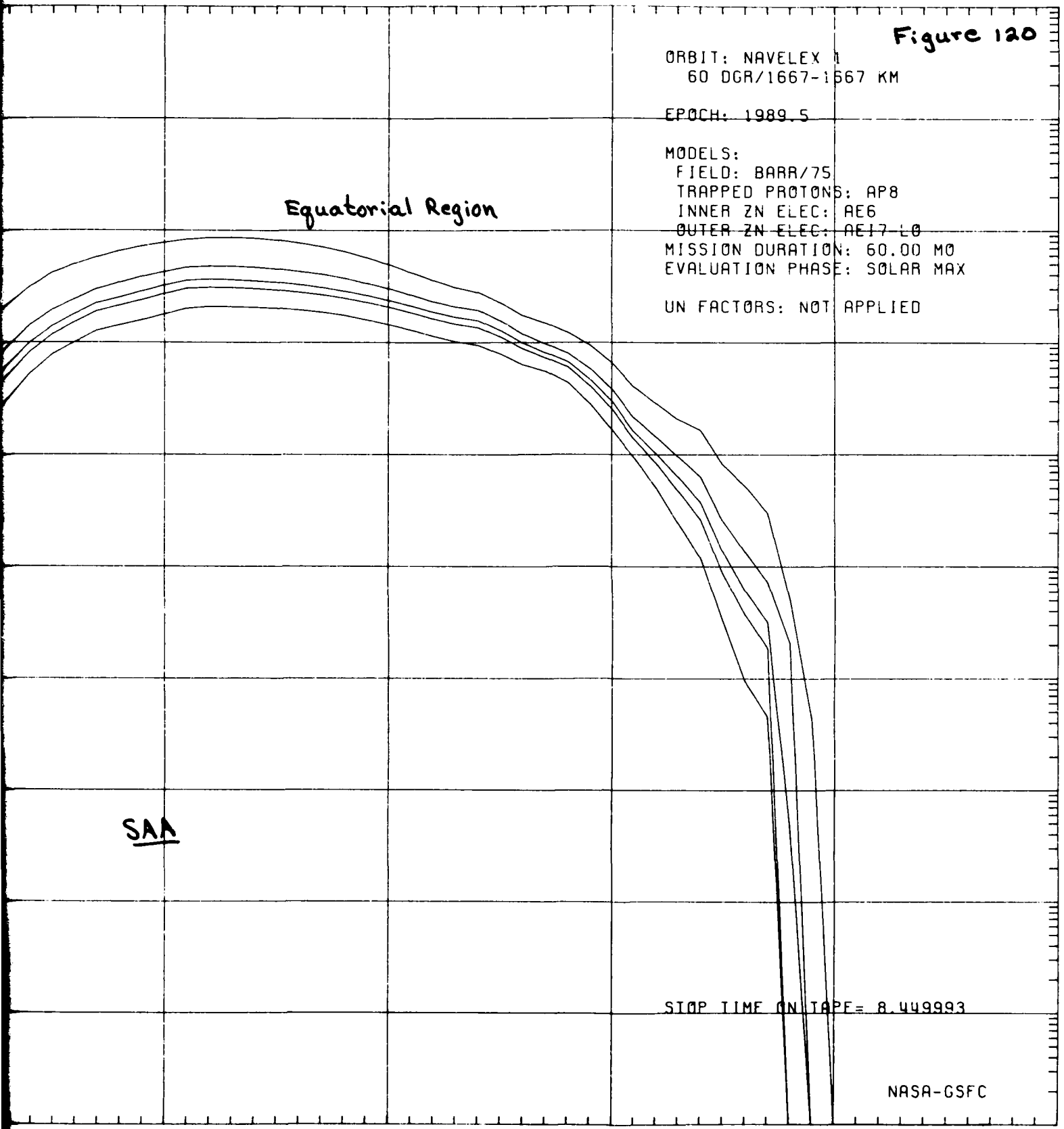
80.0

90.0

100.0

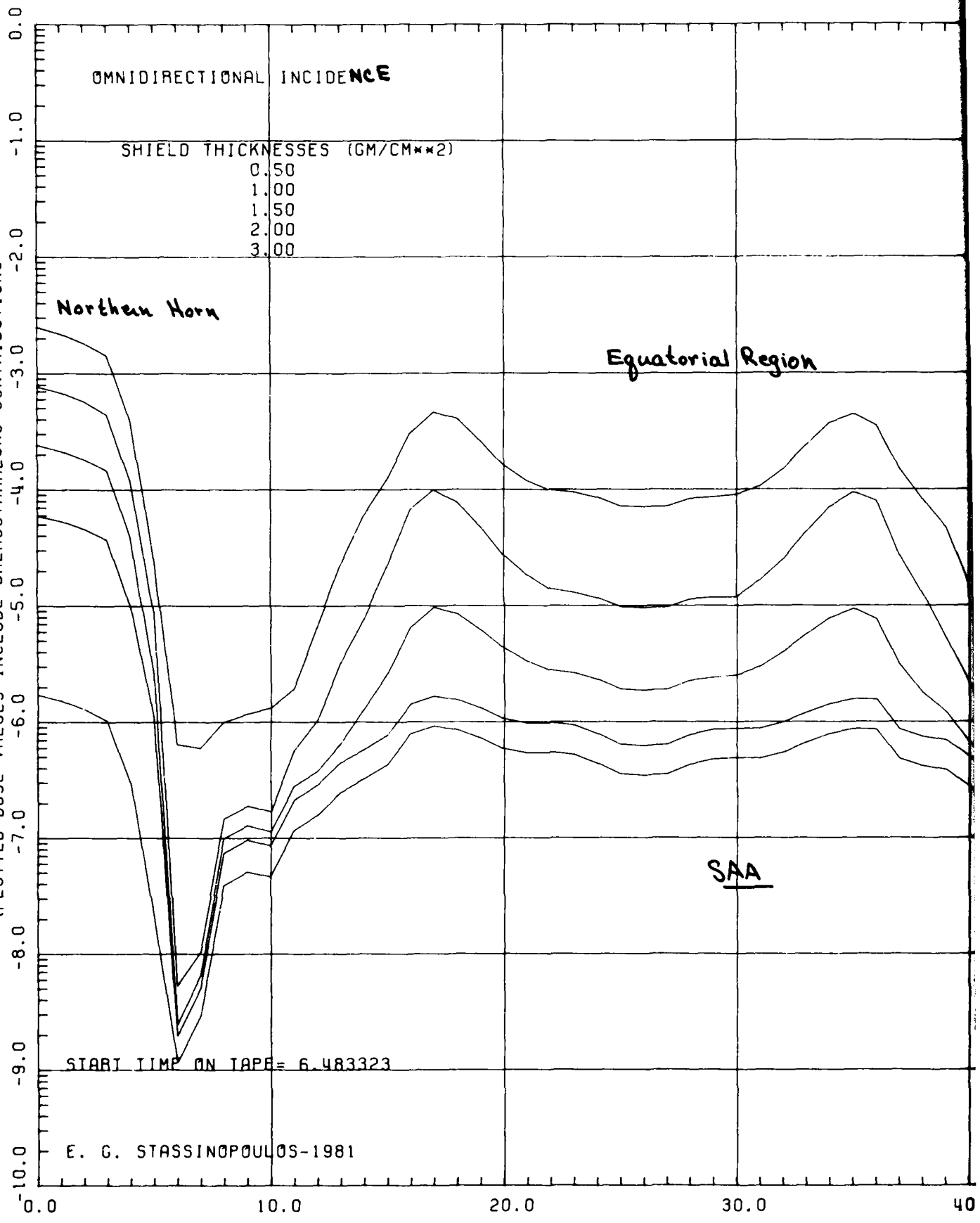
110.0

120.0



INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



OMNIDIRECTIONAL INCIDENCE

SHIELD THICKNESSES (GM/CM\*\*2)

- 0.50
- 1.00
- 1.50
- 2.00
- 3.00

Northern Horn

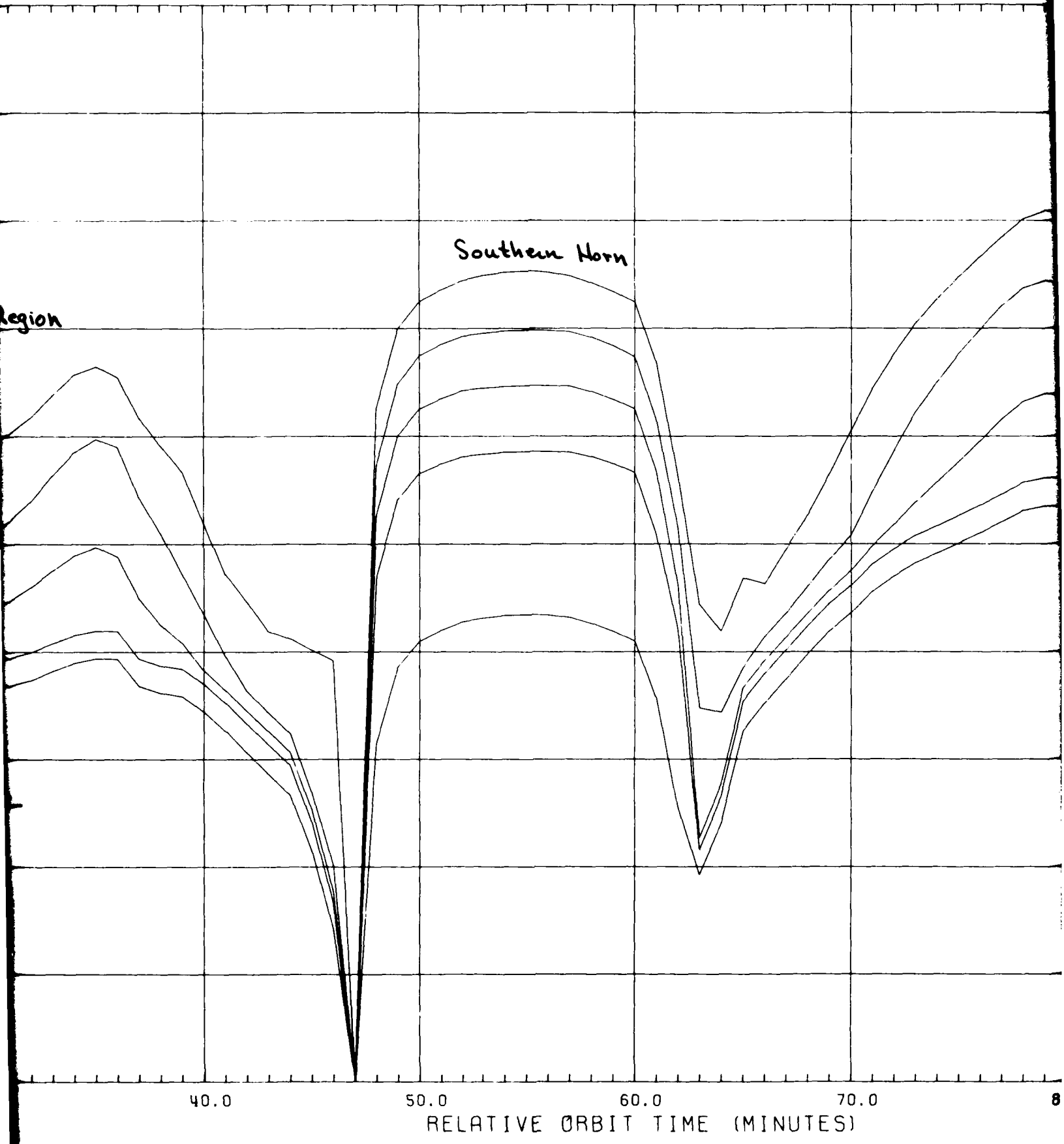
Equatorial Region

SAA

START TIME ON TAPE = 6.483323

E. G. STASSINOPOULOS-1981

DOSE AT CENTER OF ALUMINUM SPHERES



Region

Southern Horn

Figure 121

ORBIT: NAVELEX 1  
60 DGR/1667-1567 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

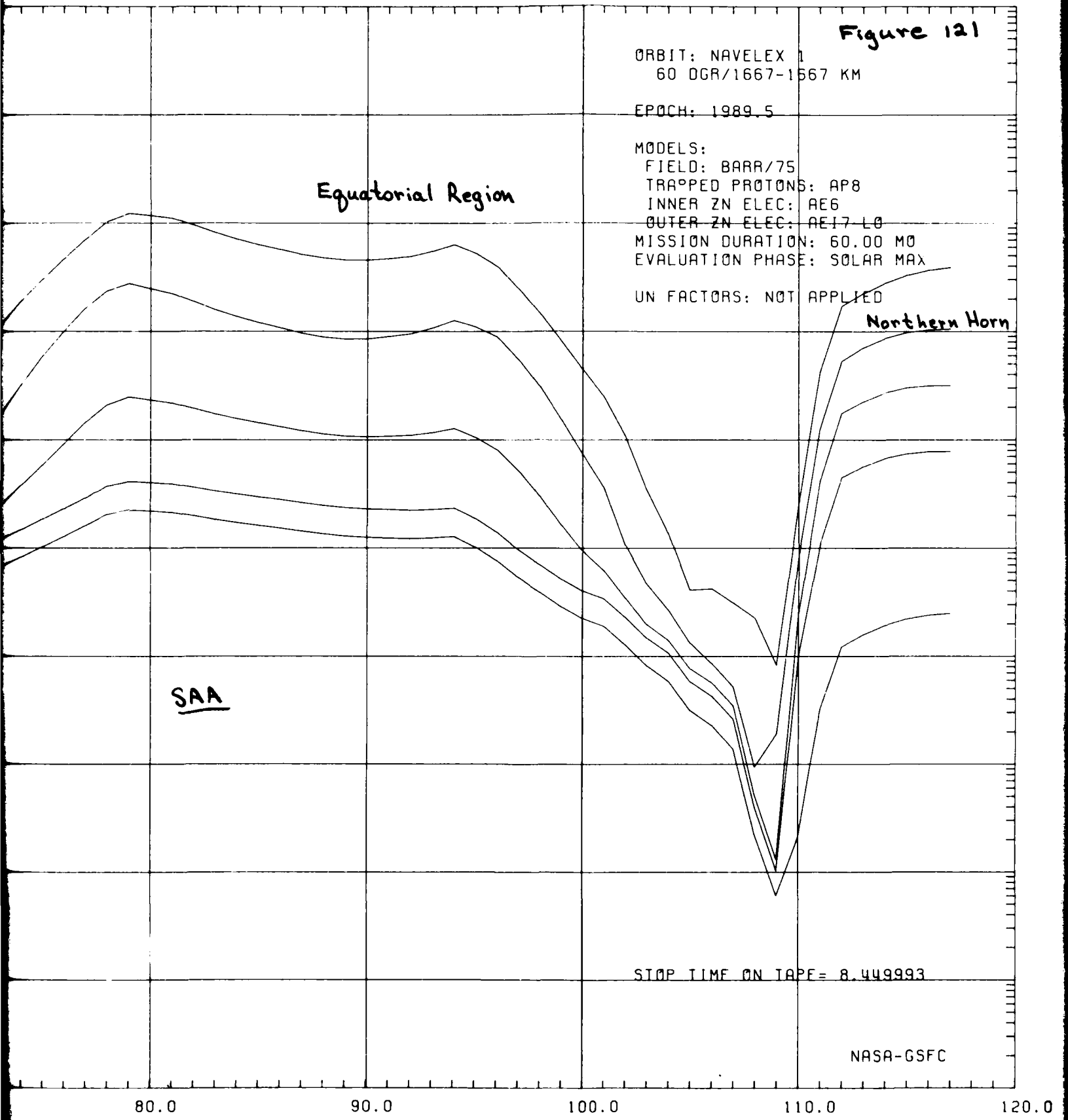
Equatorial Region

Northern Horn

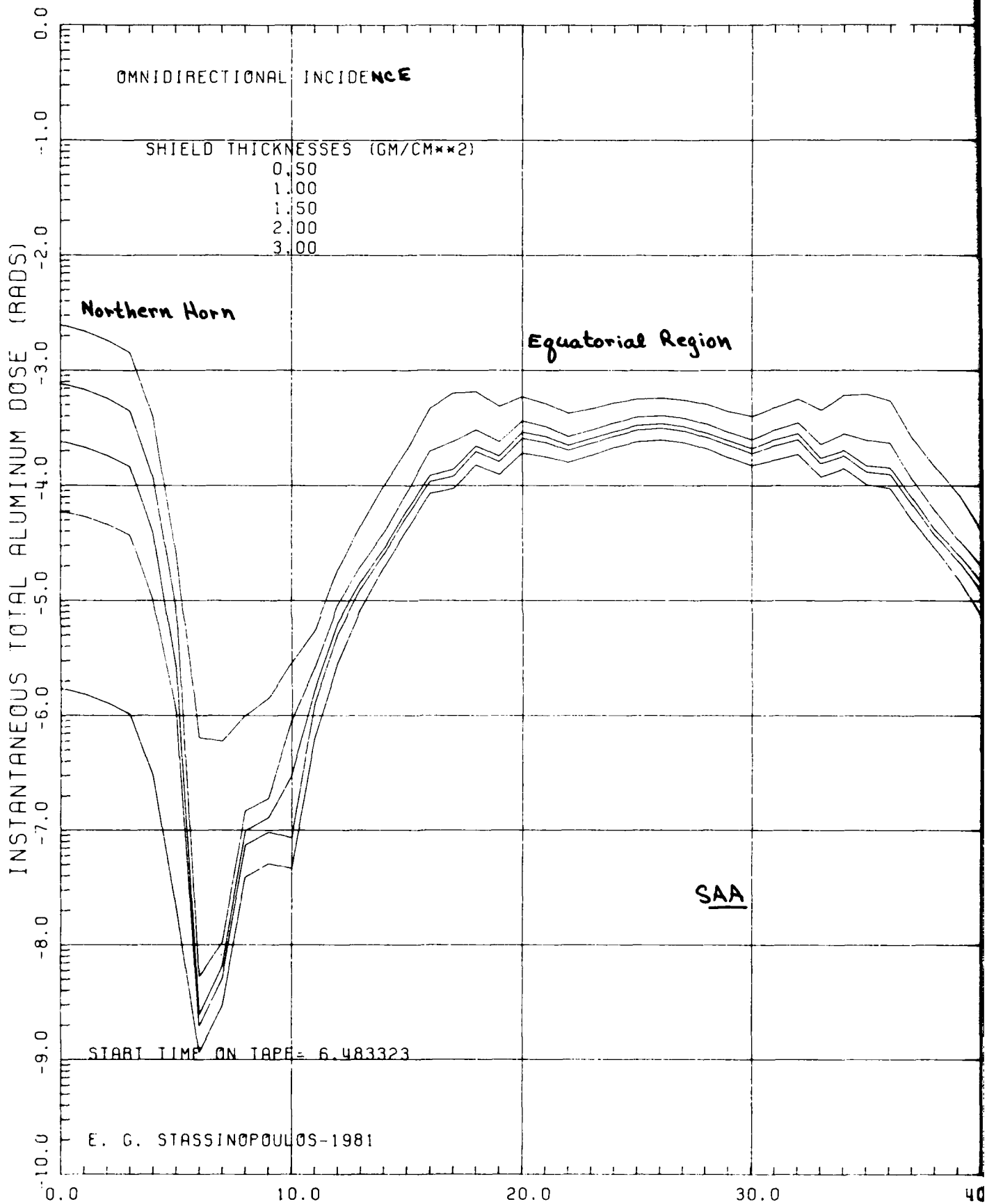
SAA

STOP TIME ON TAPE = 8.449993

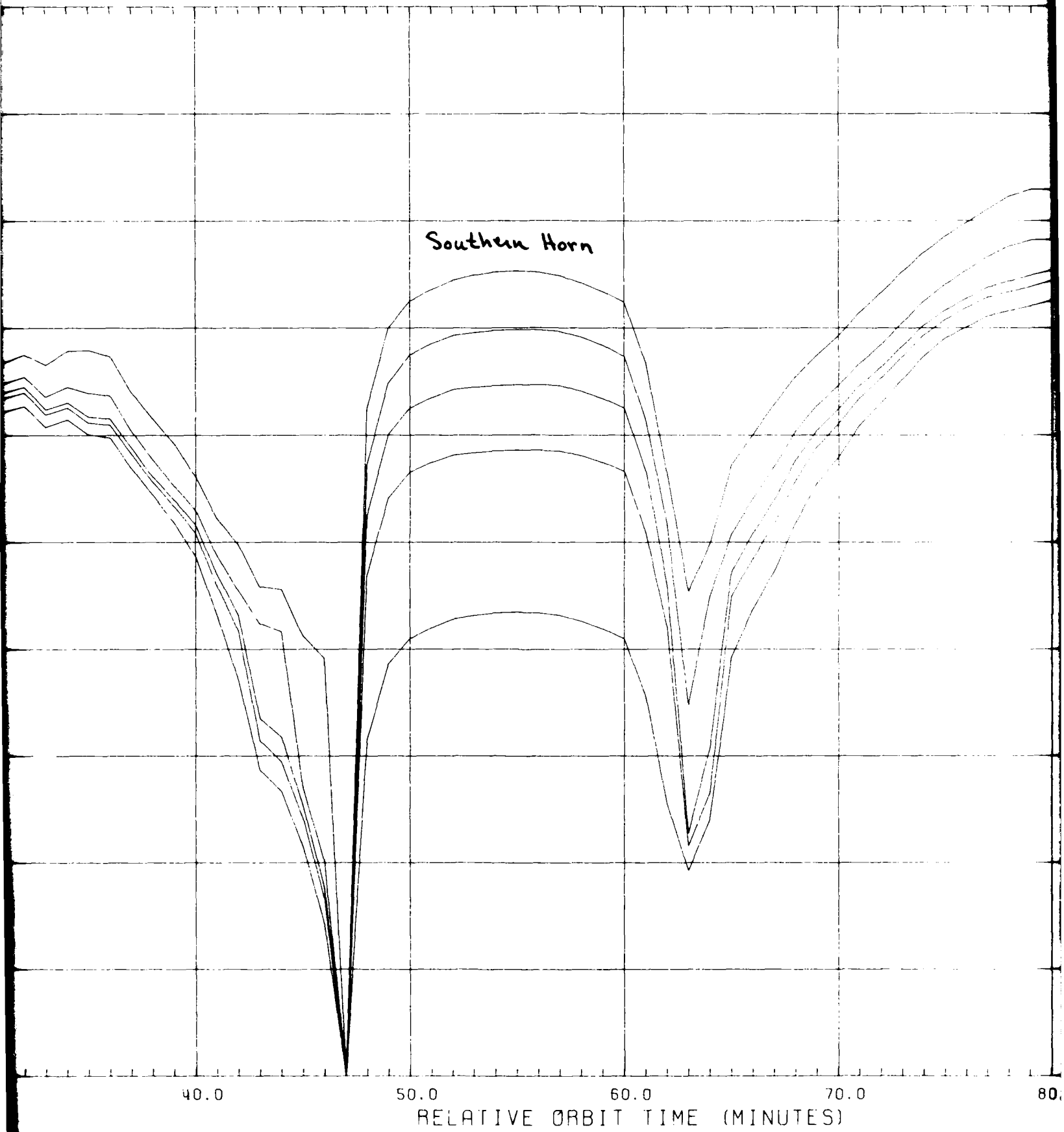
NASA-GSFC







DOSE AT CENTER OF ALUMINUM SPHERES



Southern Horn

40.0

50.0

60.0

70.0

80.0

RELATIVE ORBIT TIME (MINUTES)

S

Figure 122

ORBIT: NAVELEX 1  
60 DGR/1667-1667 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial Region

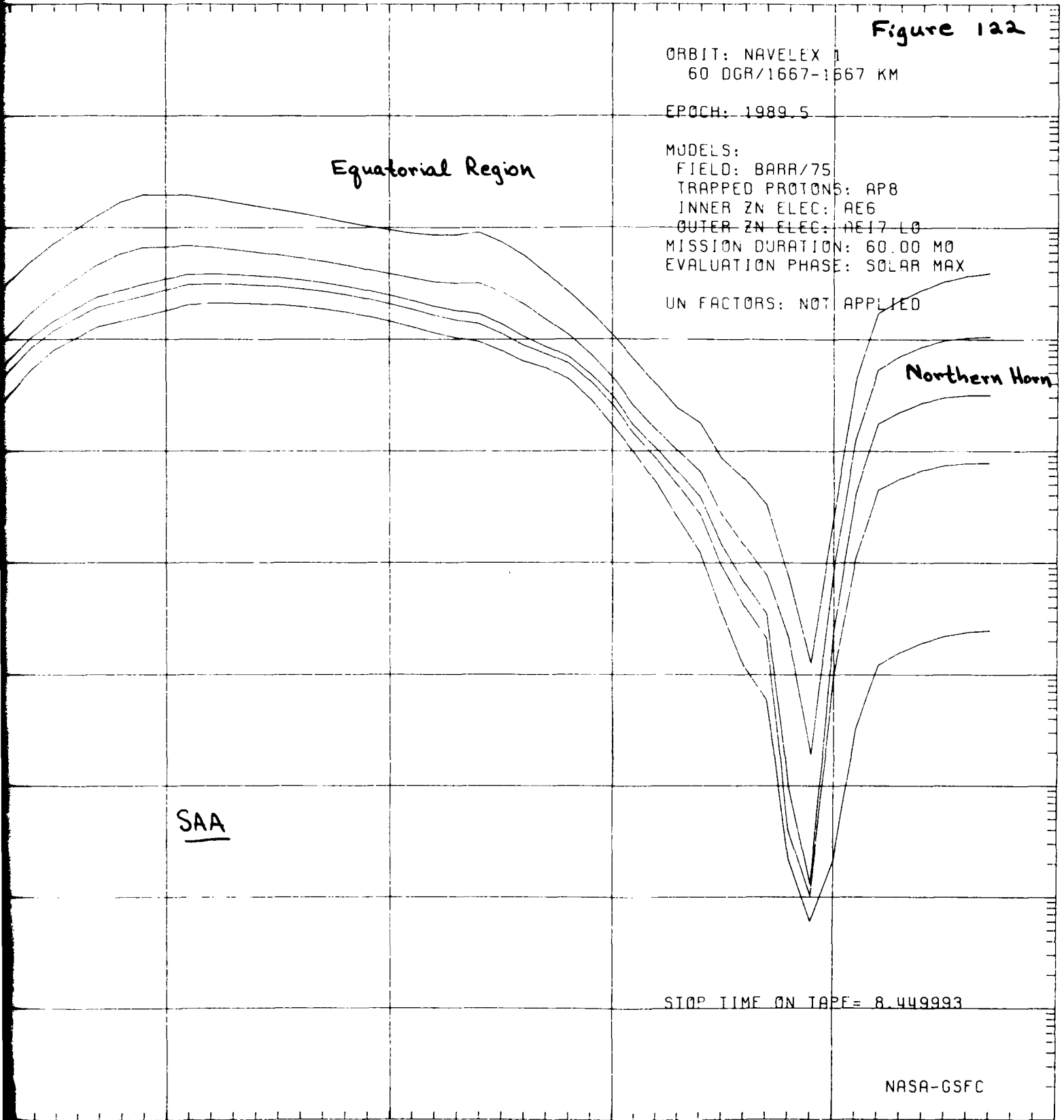
Northern Horn

SAA

STOP TIME ON TAPE = 8.449993

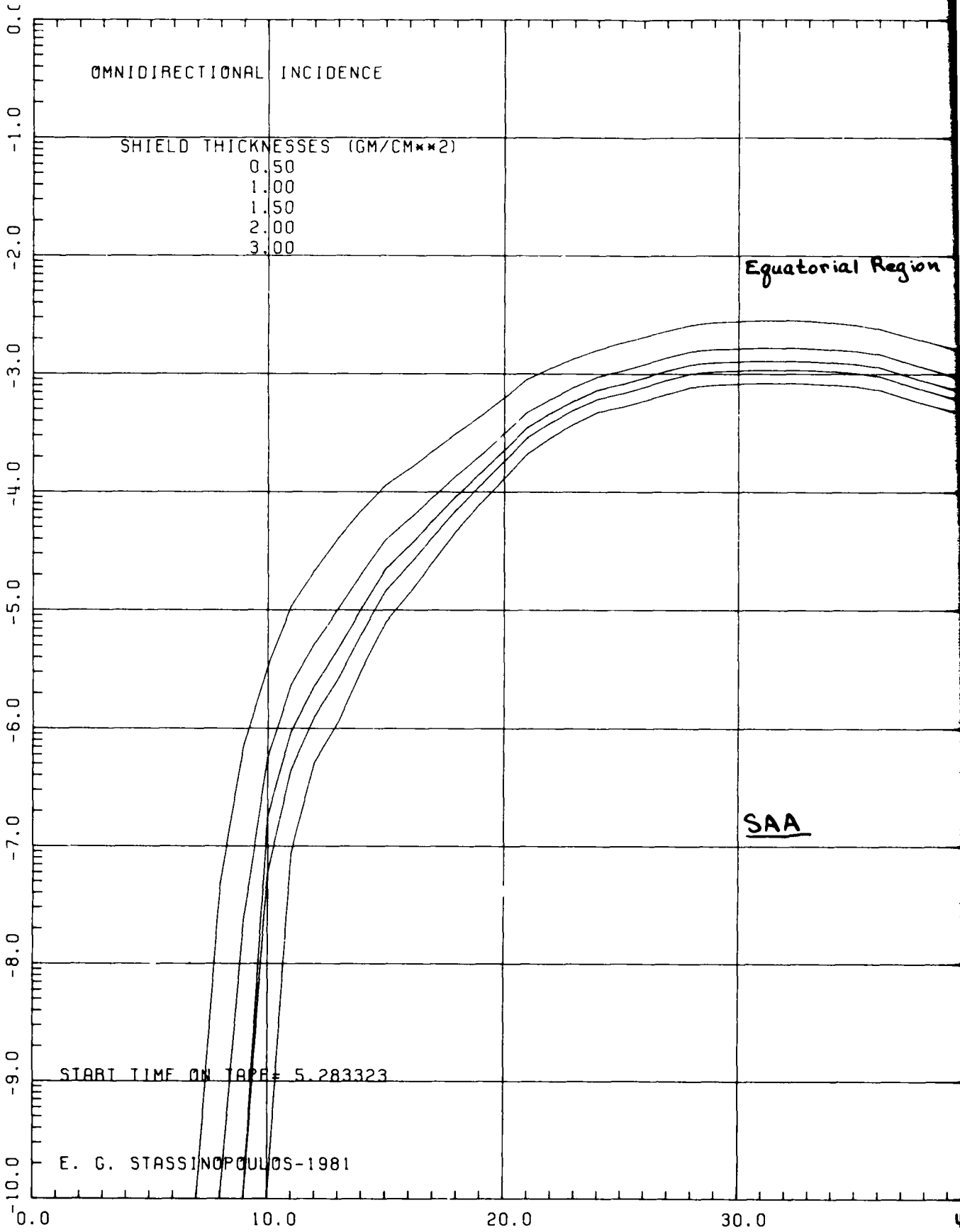
NASA-GSFC

80.0 90.0 100.0 110.0 120.0



INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

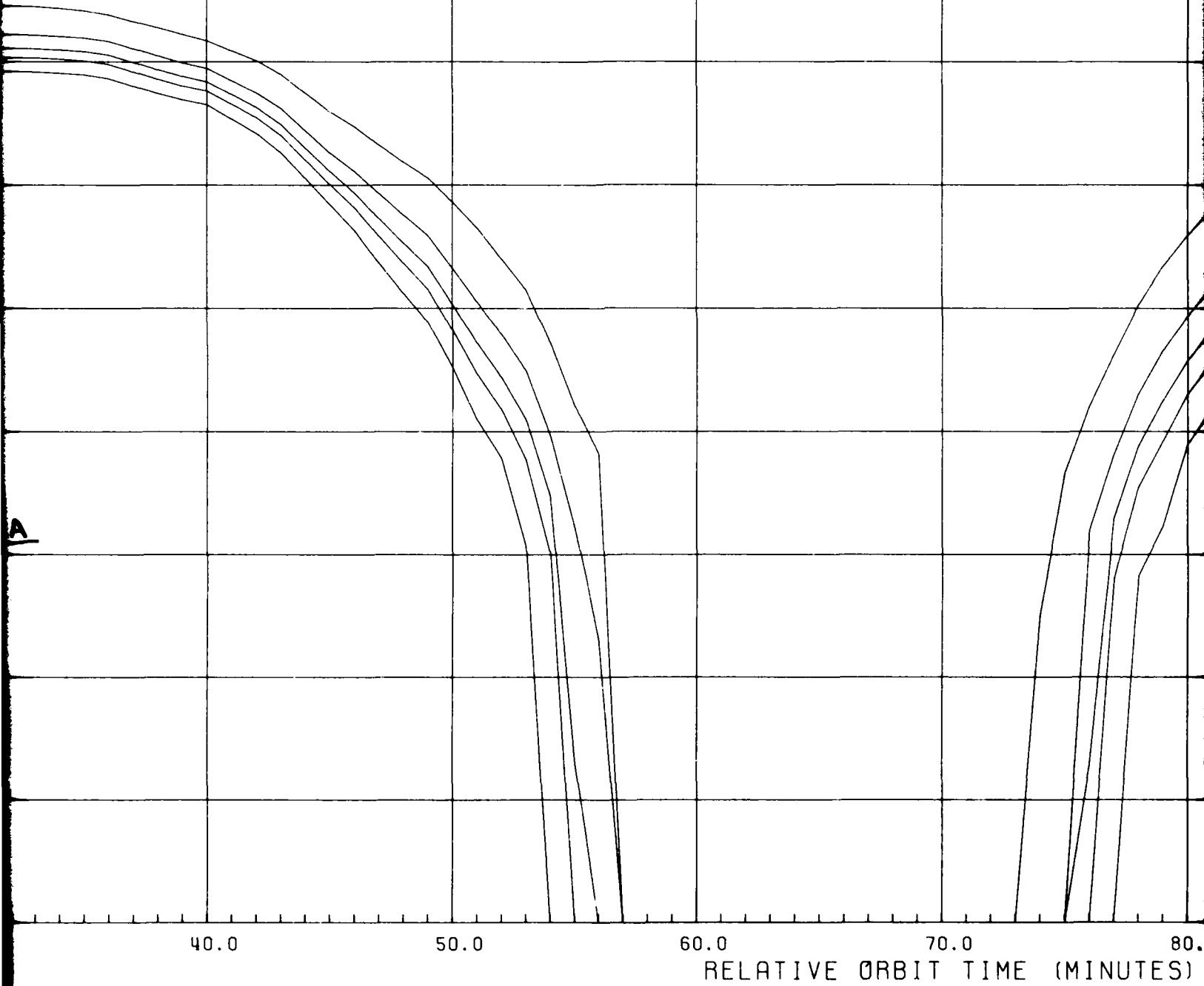
(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



DOSE AT TRANSMISSION SURFACE OF FINITE AL

( = DOSE IN SEMI-INFINITE ALUMINUM

atorial Region



3

OF FINITE ALUMINUM SLAB SHIELDS

FINITE ALUMINUM MEDIUM)

Equatorial Region

SAA

80.0

90.0

100.0

110.0

120.0

(MINUTES)

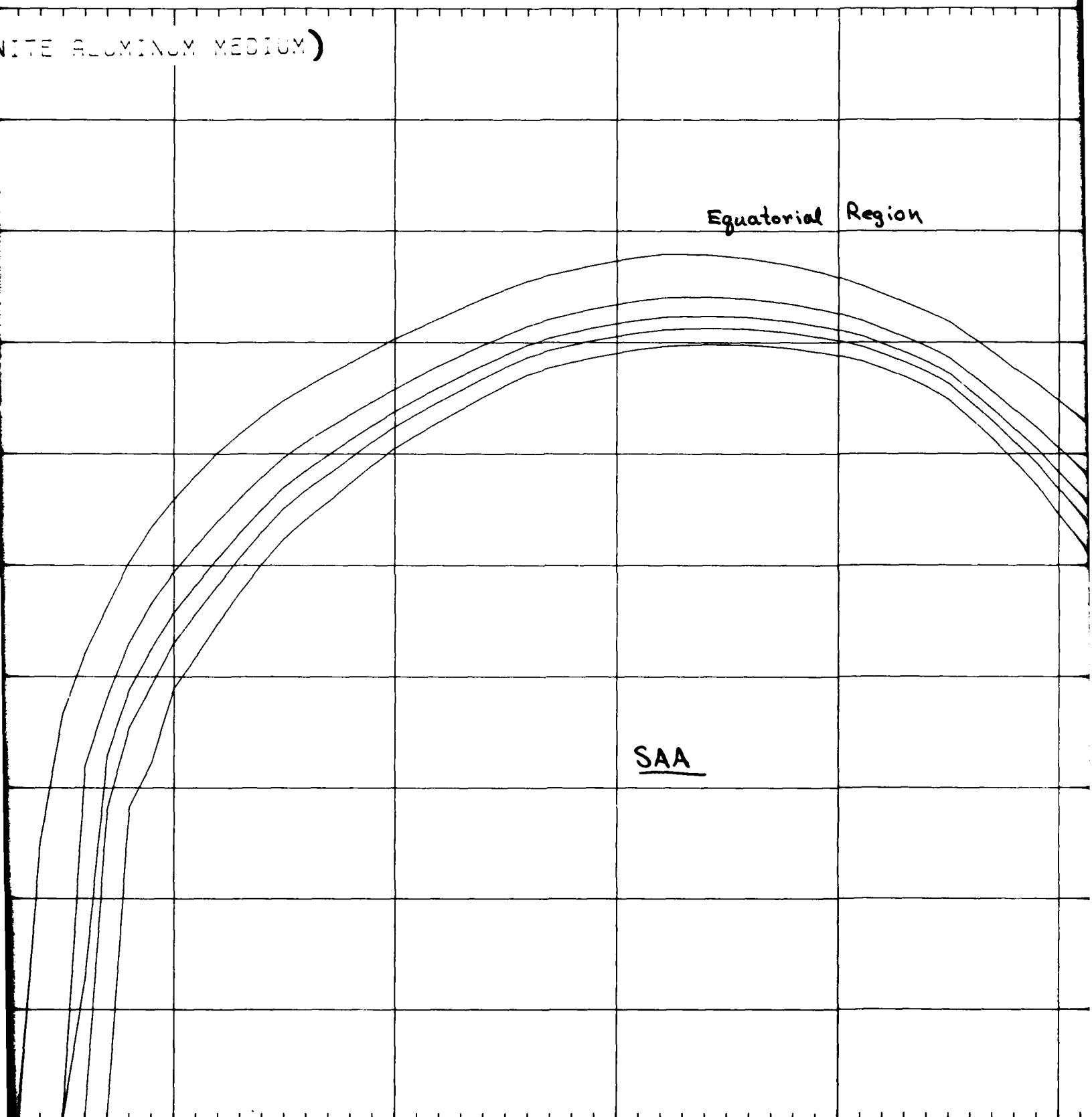


Figure 123

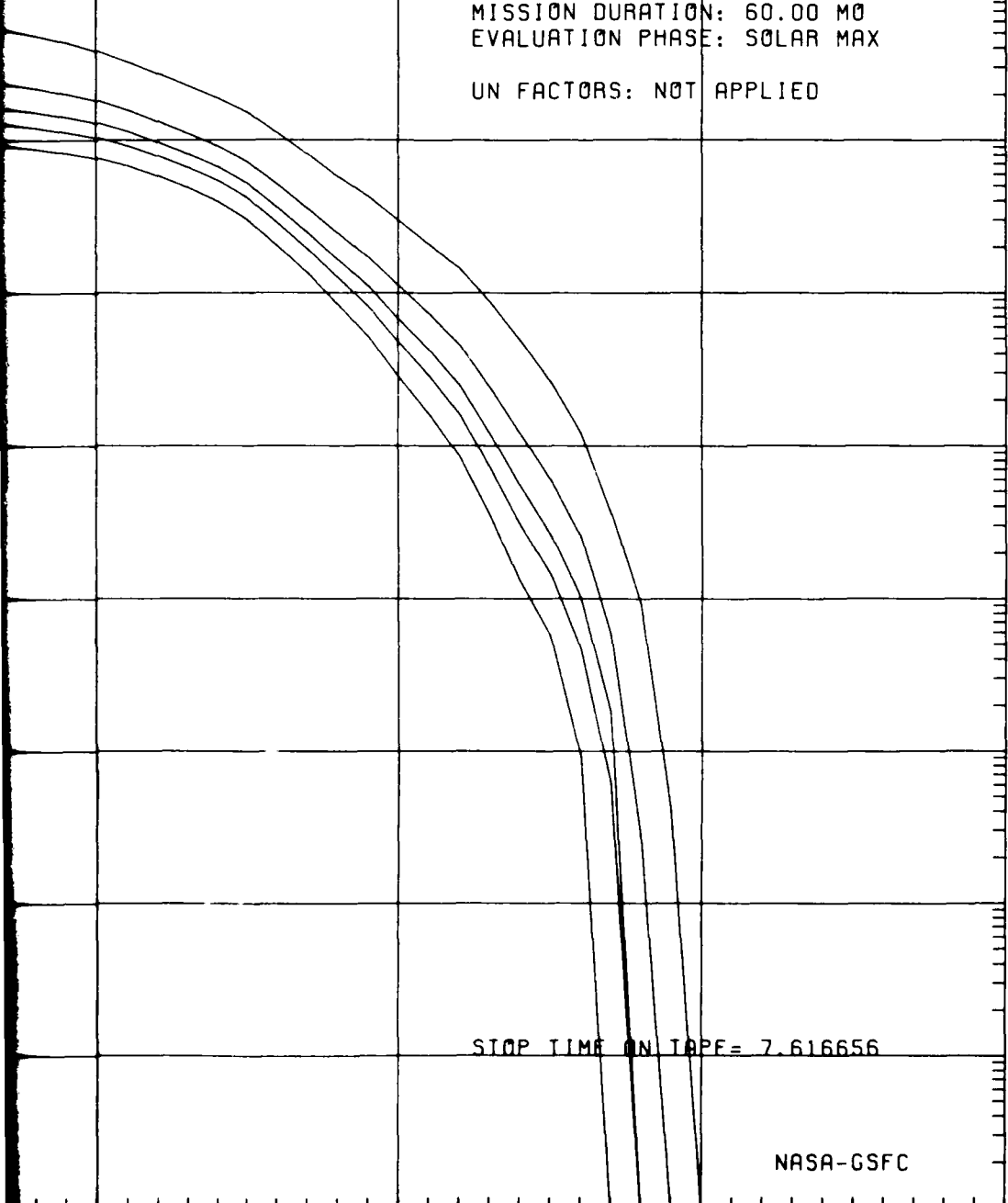
ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Orbital Region



STOP TIME ON TAPE = 7.616656

NASA-GSFC

110.0

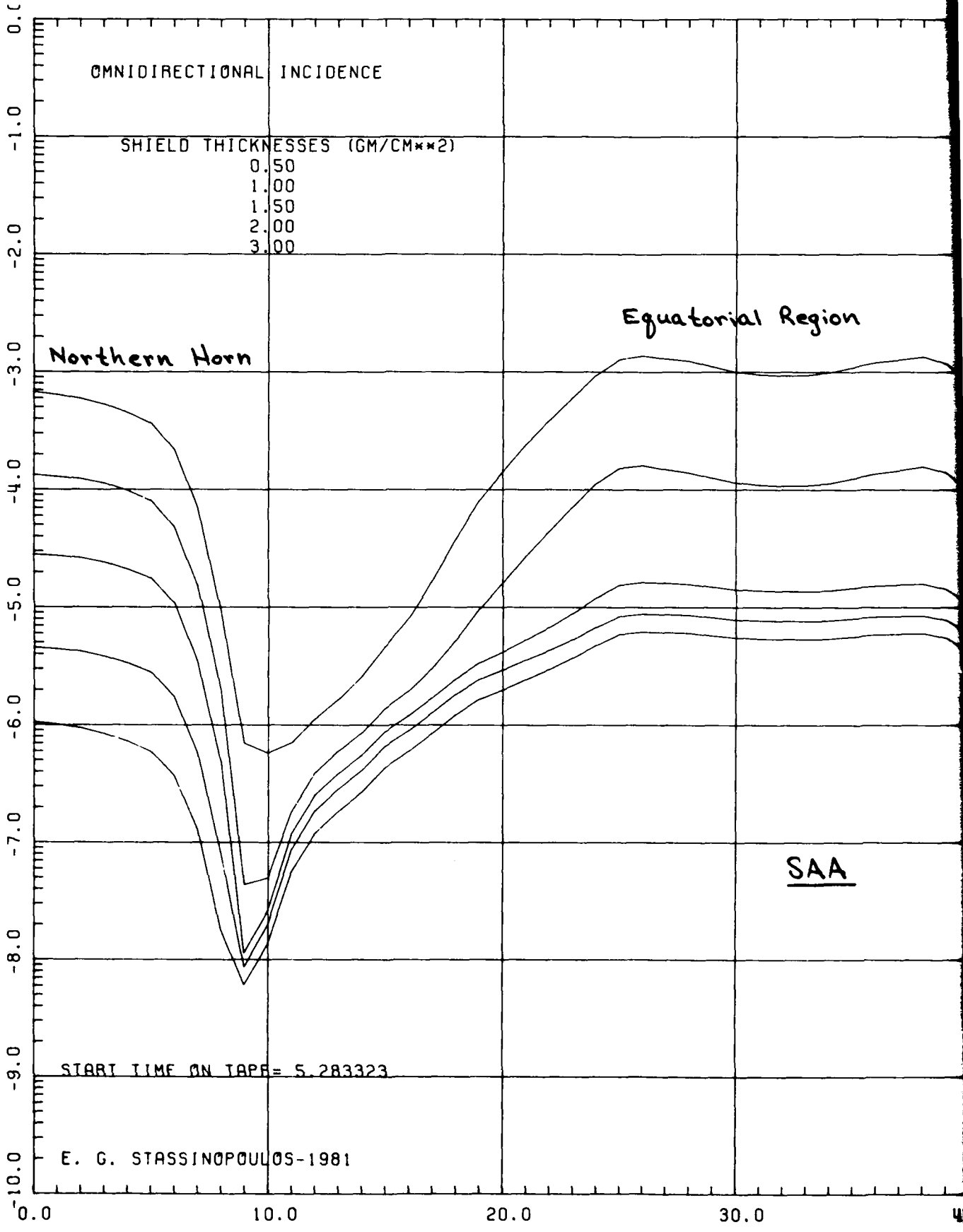
120.0

130.0

140.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

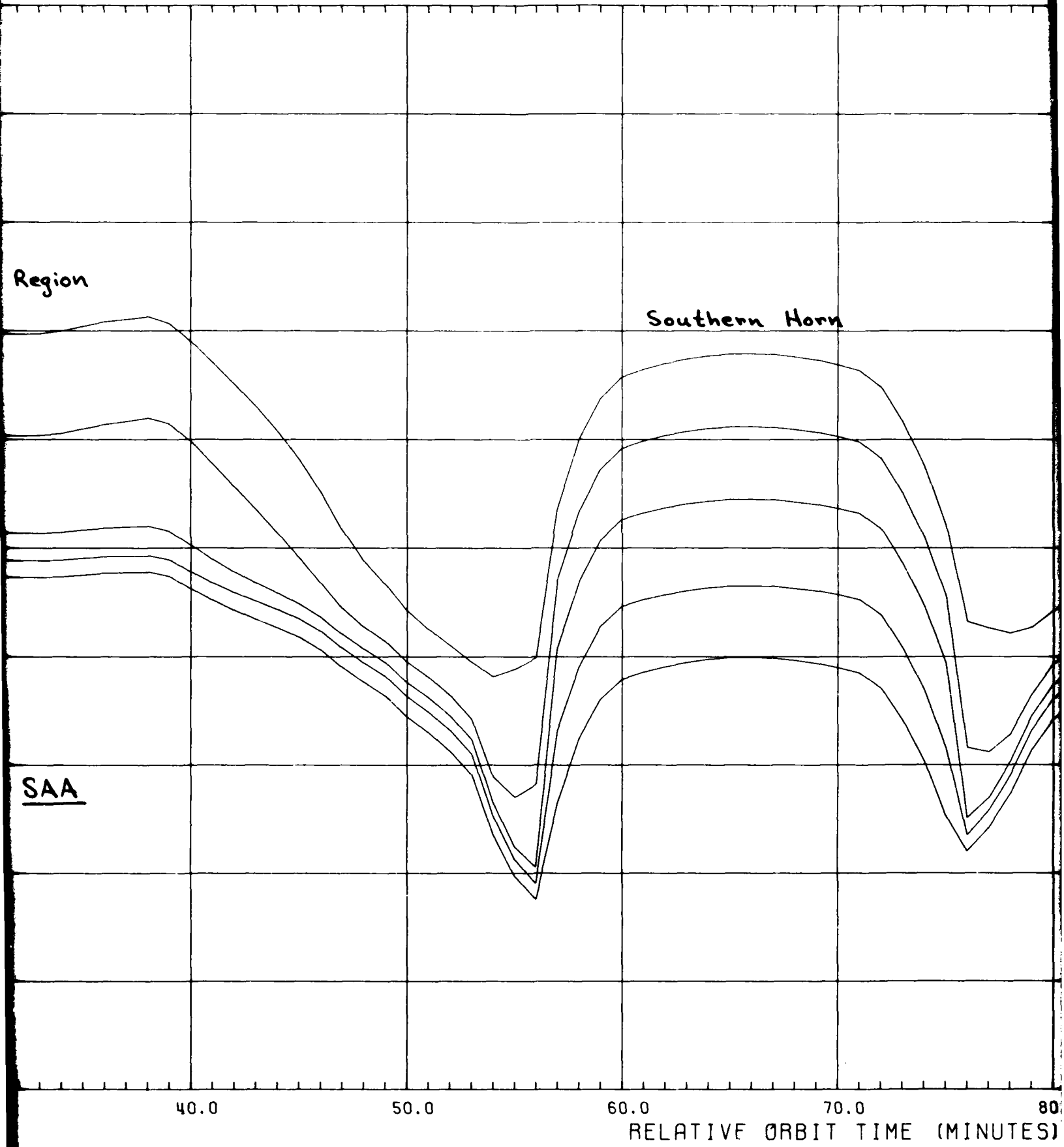
(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)





2

DOSE AT TRANSMISSION SURFACE OF FINITE R



3'

# OF FINITE ALUMINUM SLAB SHIELDS

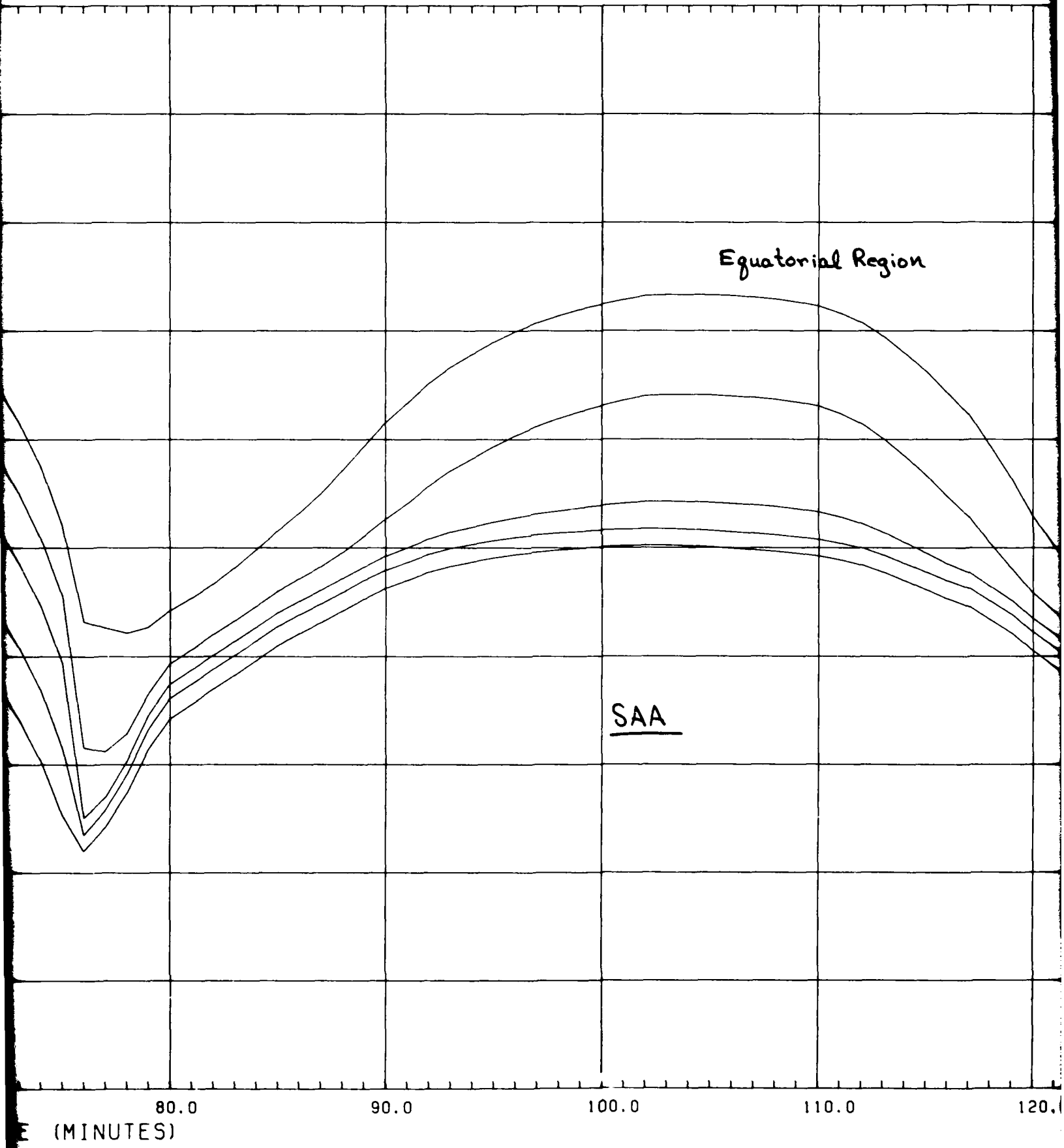


Figure 124

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

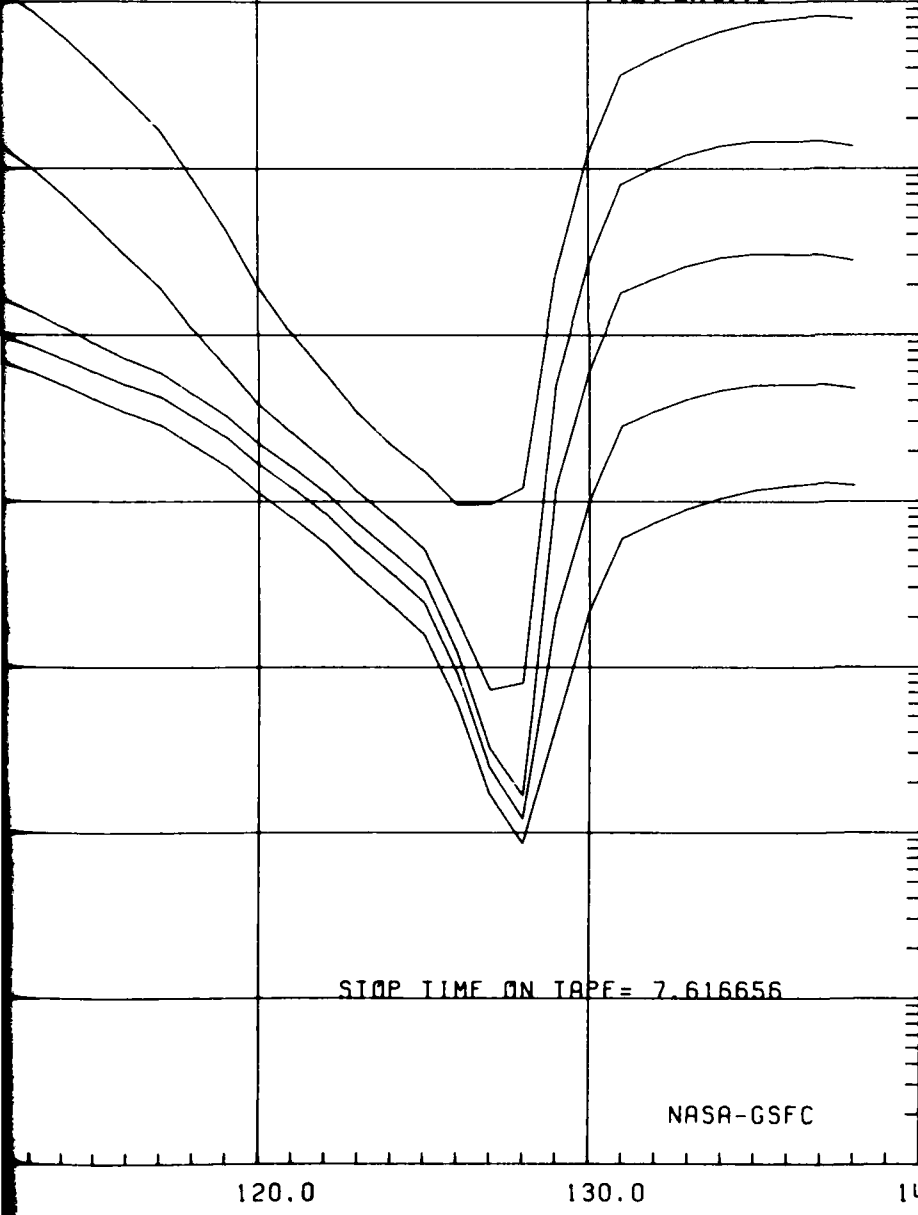
MODELS:

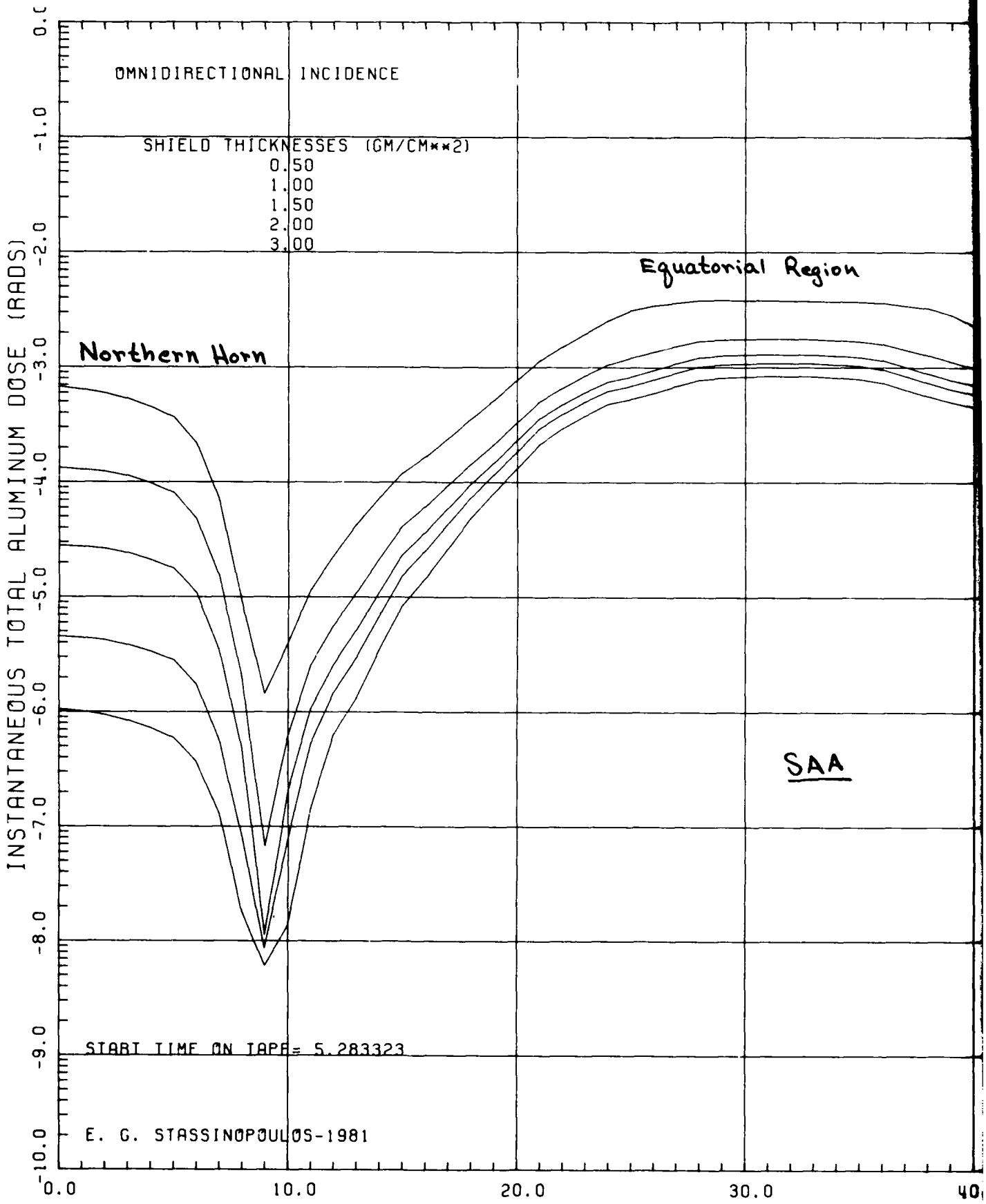
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn

Region





DOSE AT TRANSMISSION SURFACE OF FINITE P

Region

Southern Horn

SAA

40.0

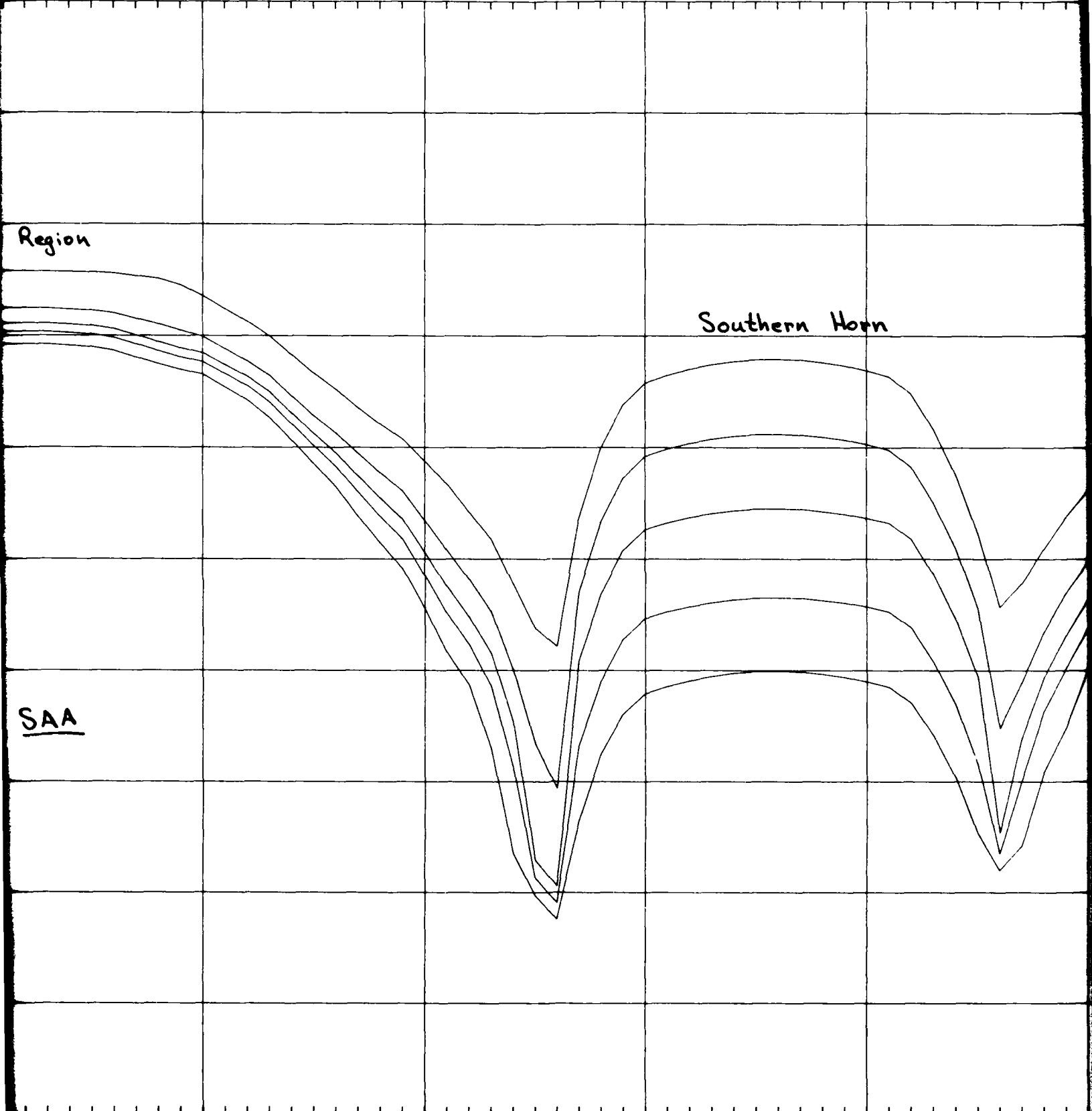
50.0

60.0

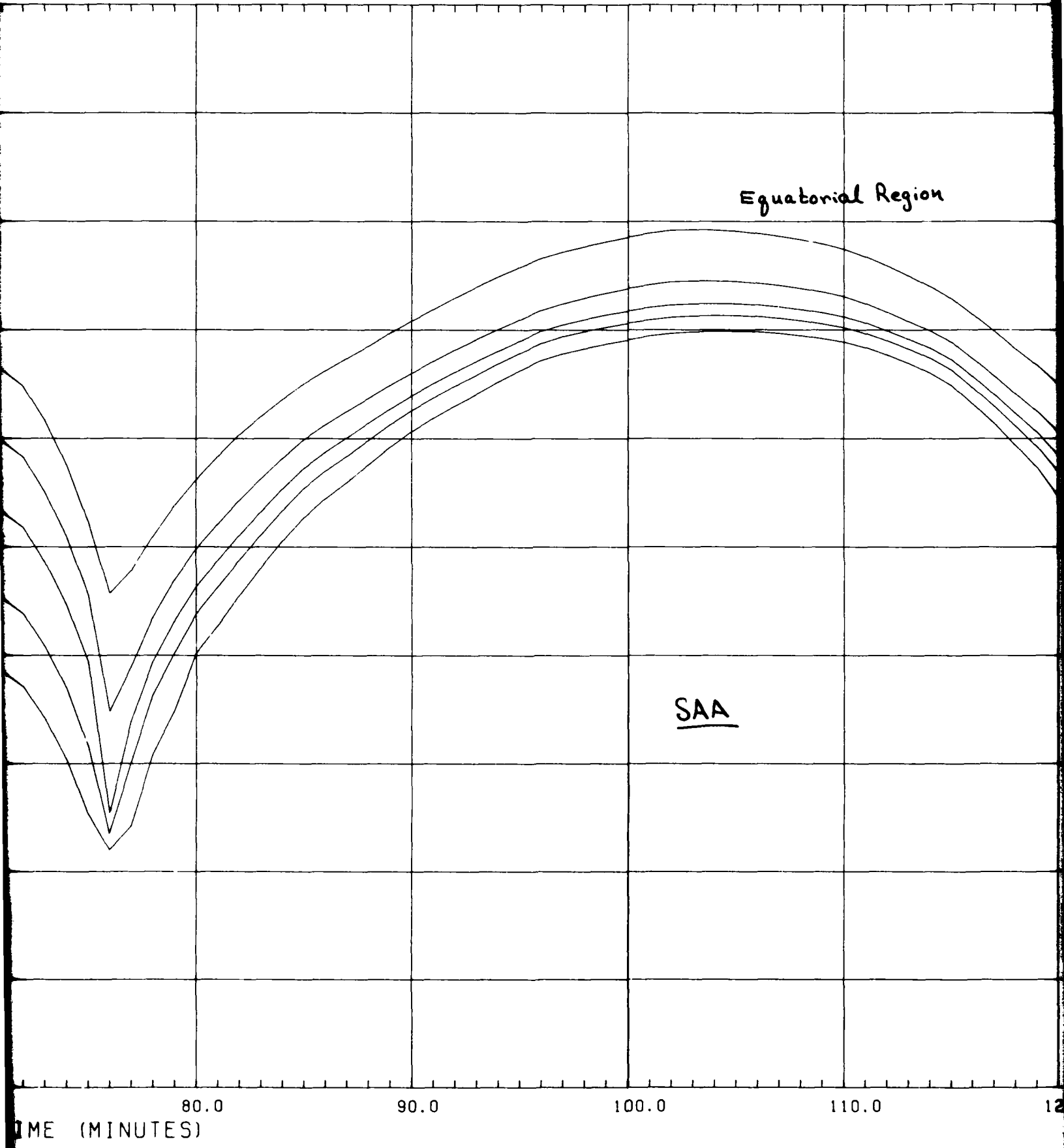
70.0

80.0

RELATIVE ORBIT TIME (MINUTES)



E OF FINITE ALUMINUM SLAB SHIELDS



Equatorial Region

SAA

TIME (MINUTES) 80.0 90.0 100.0 110.0 120.0

Figure 125

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

OUTER ZN ELEC: AE17 L0

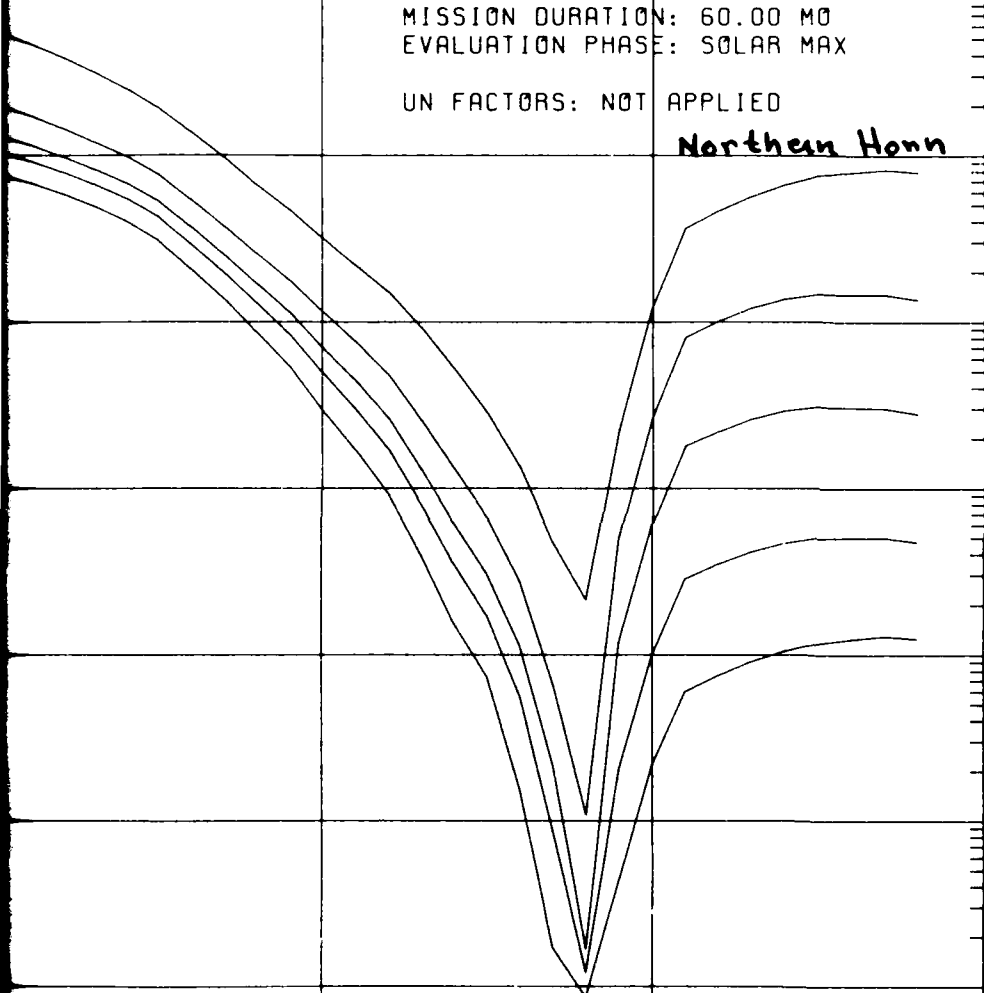
MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Hemisphere

Region



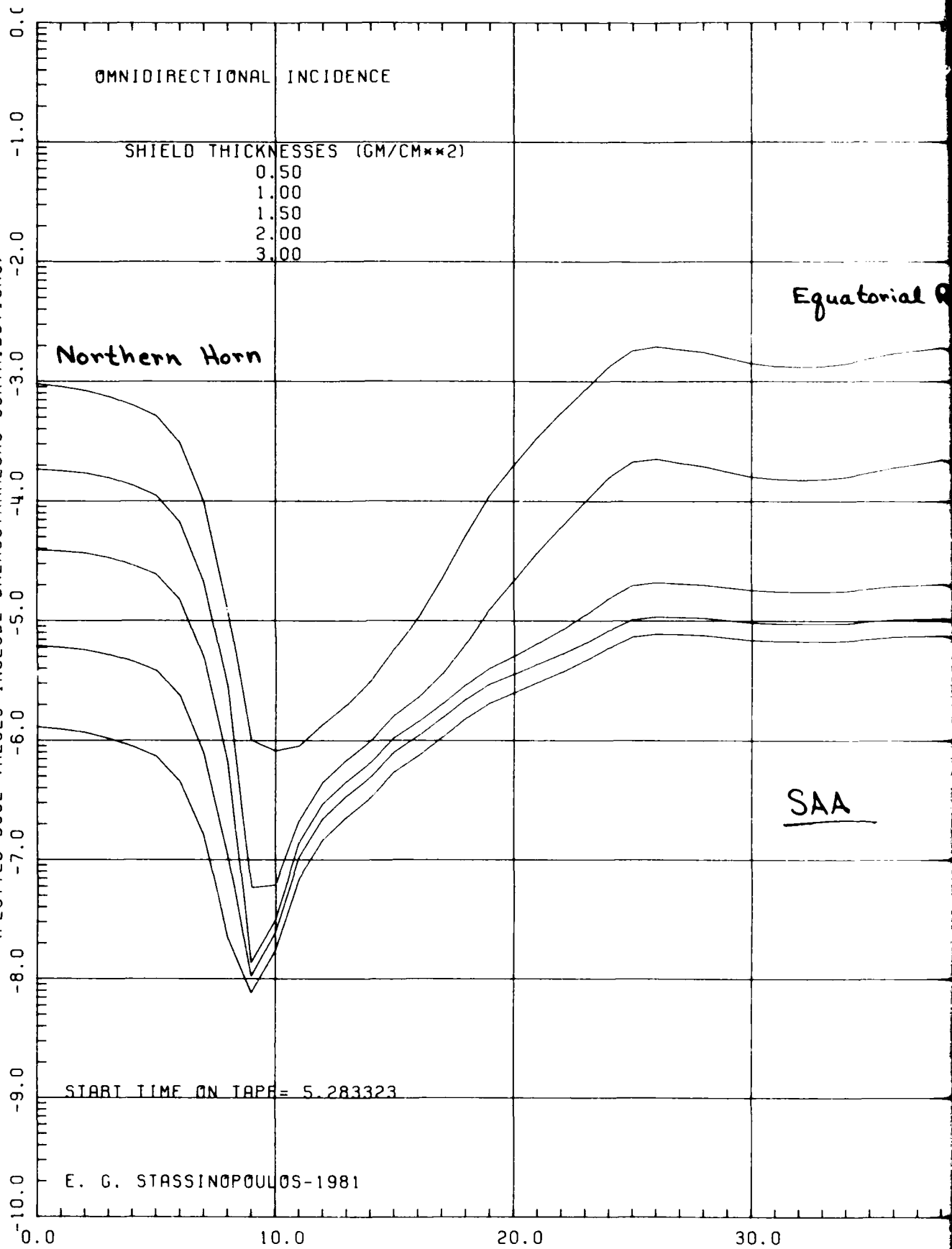
STOP TIME ON TAPE = 7.616656

NASA-GSFC

120.0 130.0 140.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)





DOSE IN SEMI-INFINITE ALUMINUM

Equatorial Region

Southern Horn

SAA

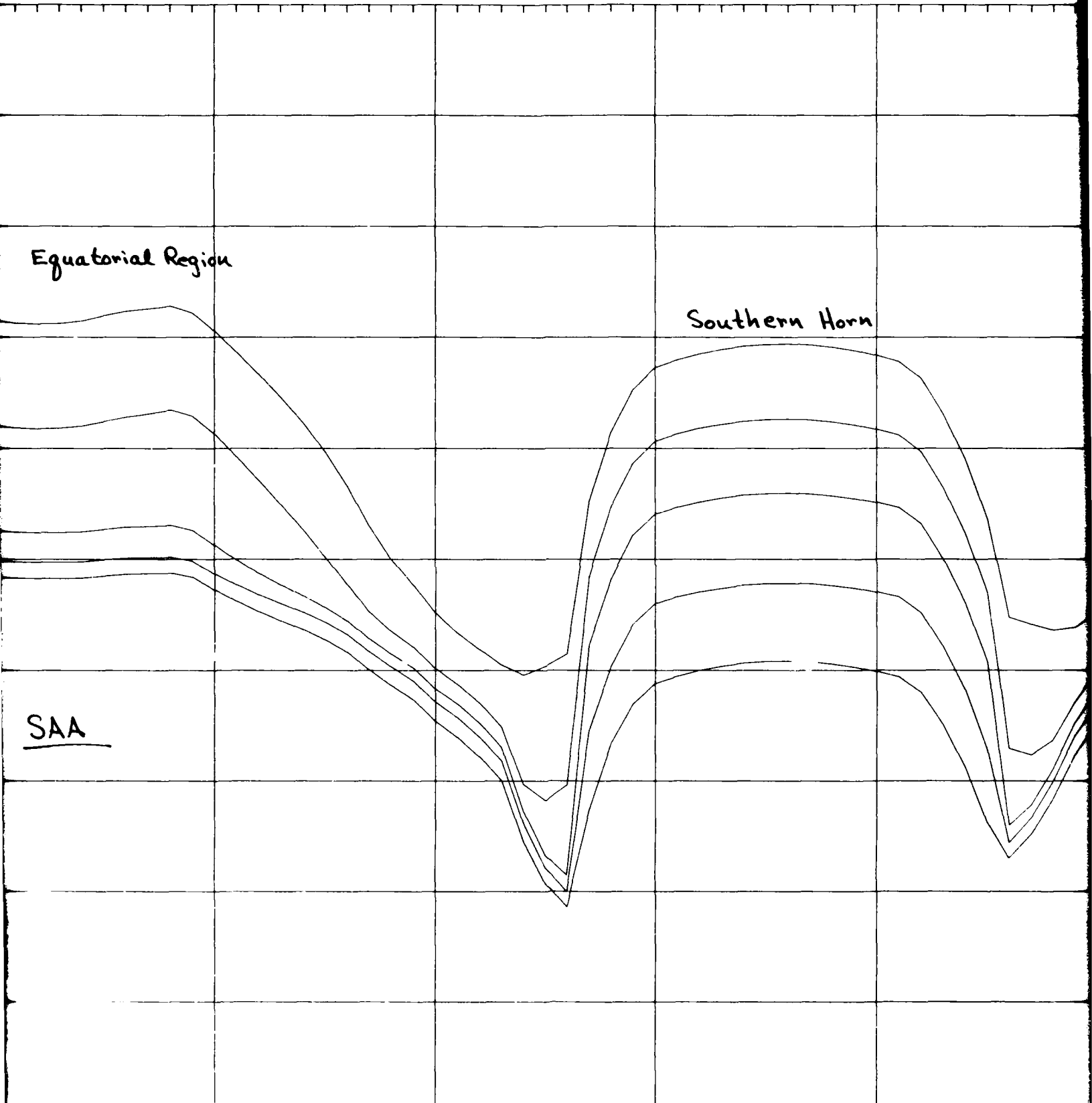
40.0

50.0

60.0

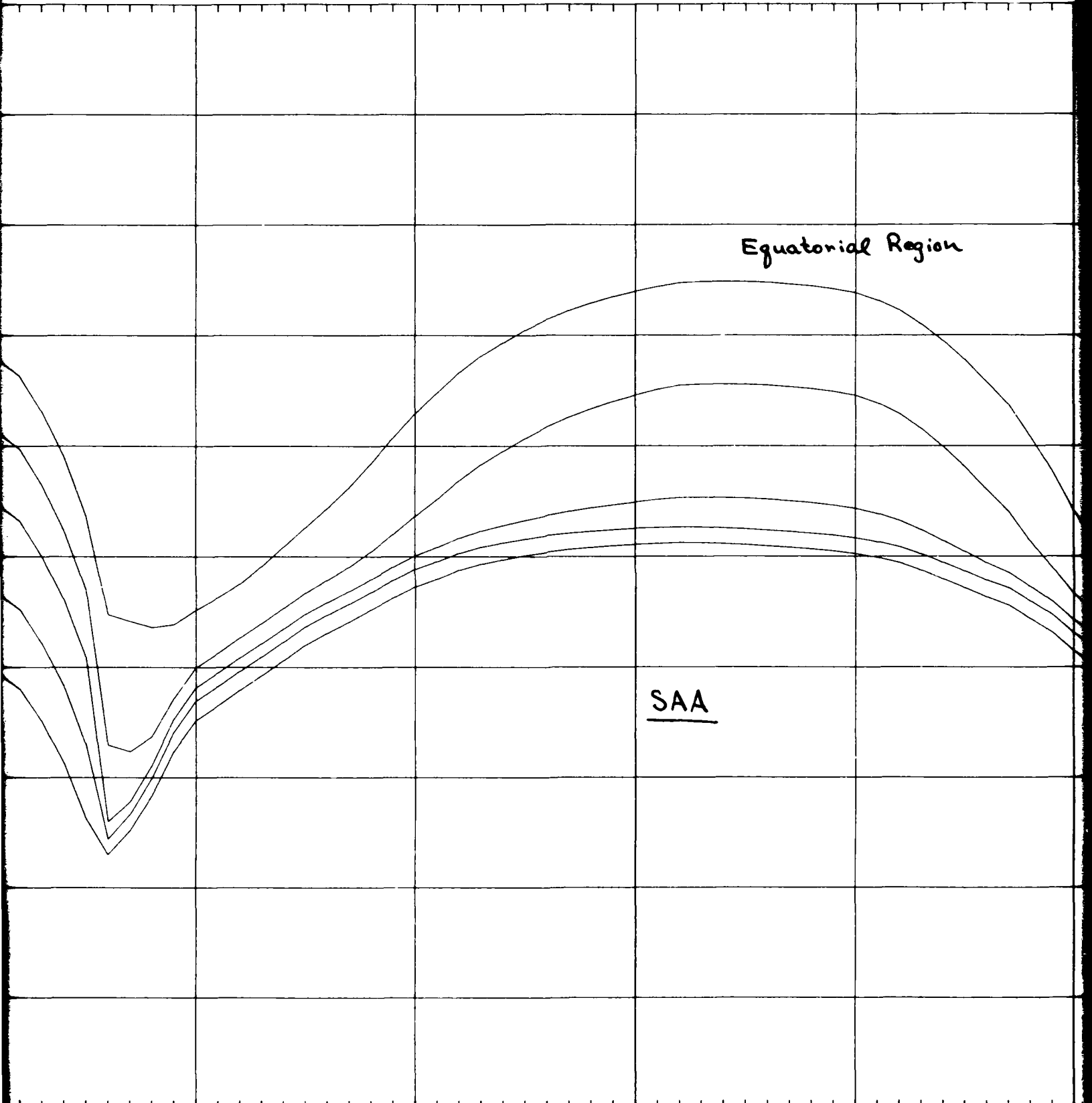
70.0

RELATIVE ORBIT TIME (MINUTES)



3

ITE ALUMINUM MEDIUM



Equatorial Region

SAA

TIME (MINUTES)

12

Figure 126

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

atorial Region

Northern  
Horn

STOP TIME ON TAPE = 7.616656

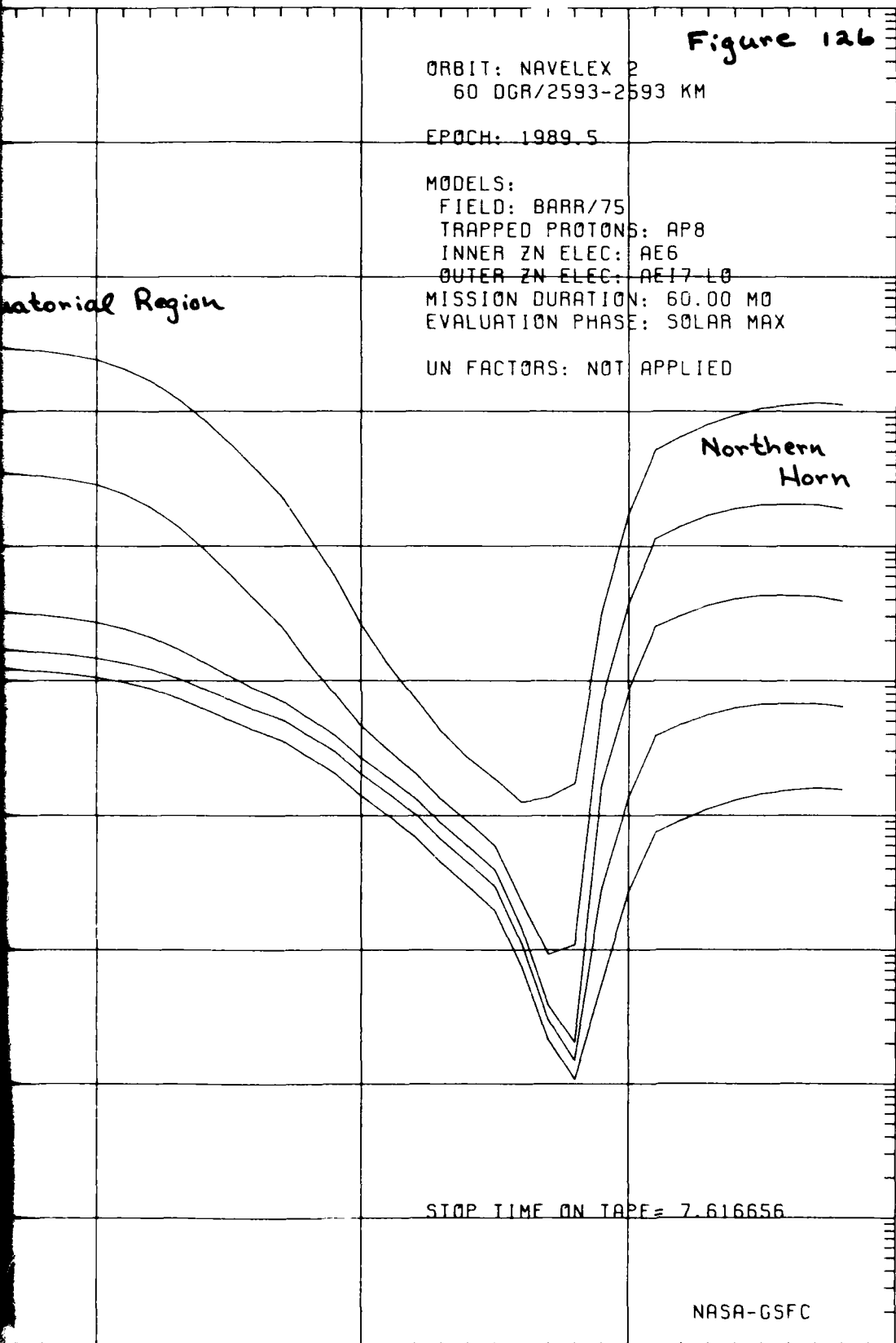
NASA-GSFC

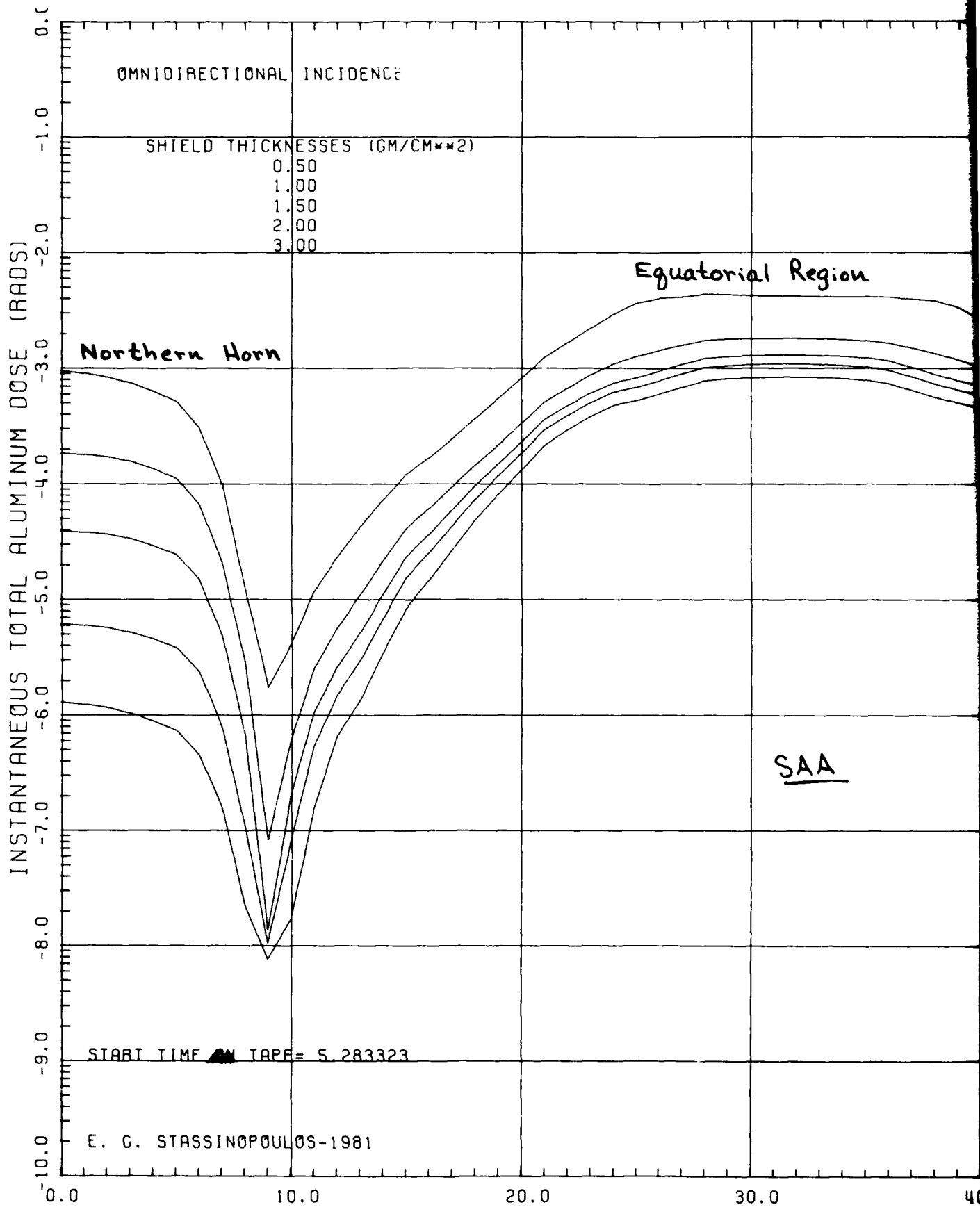
110.0

120.0

130.0

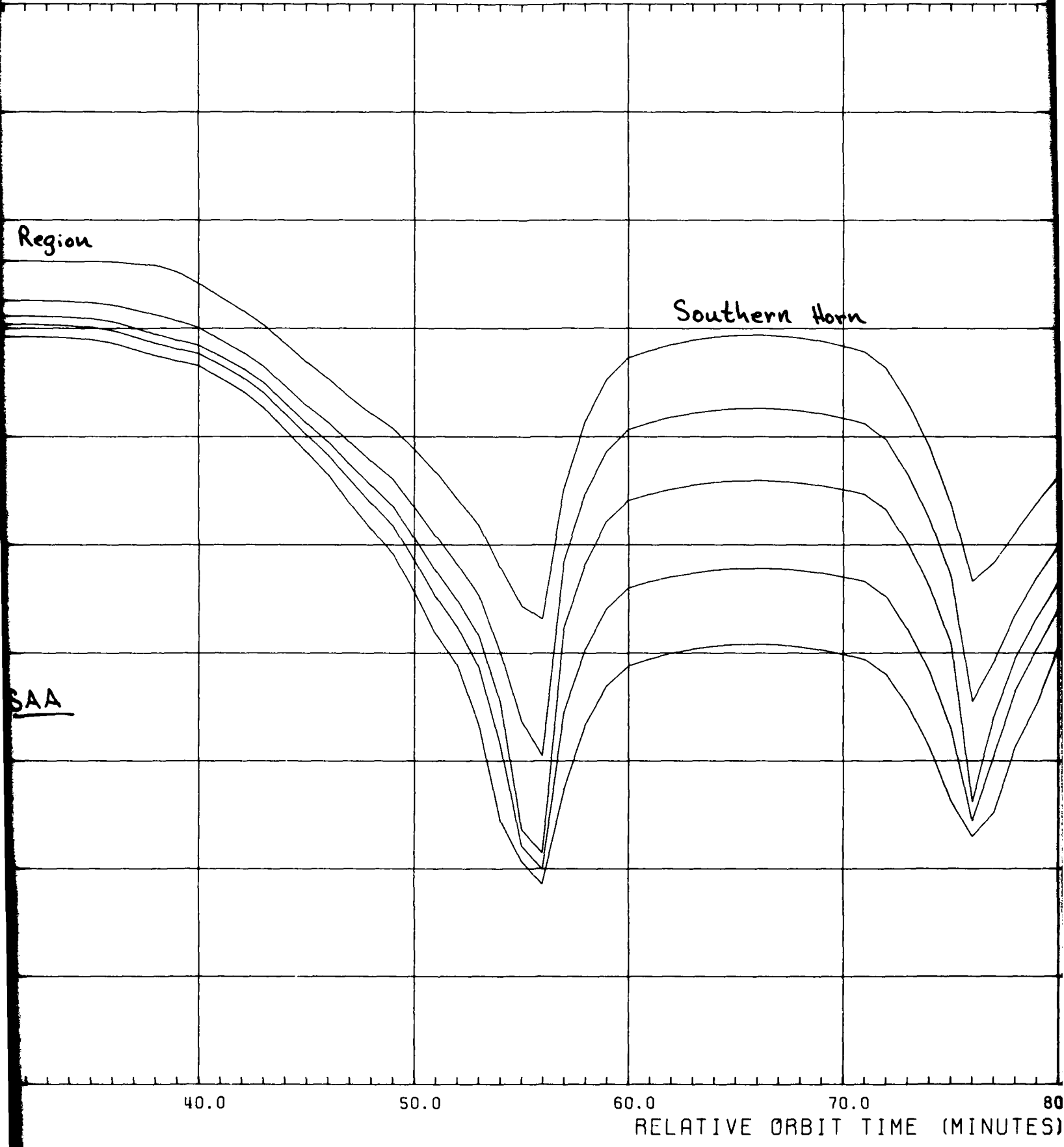
140.0





2'

DOSE IN SEMI-INFINITE ALUMINUM



Region

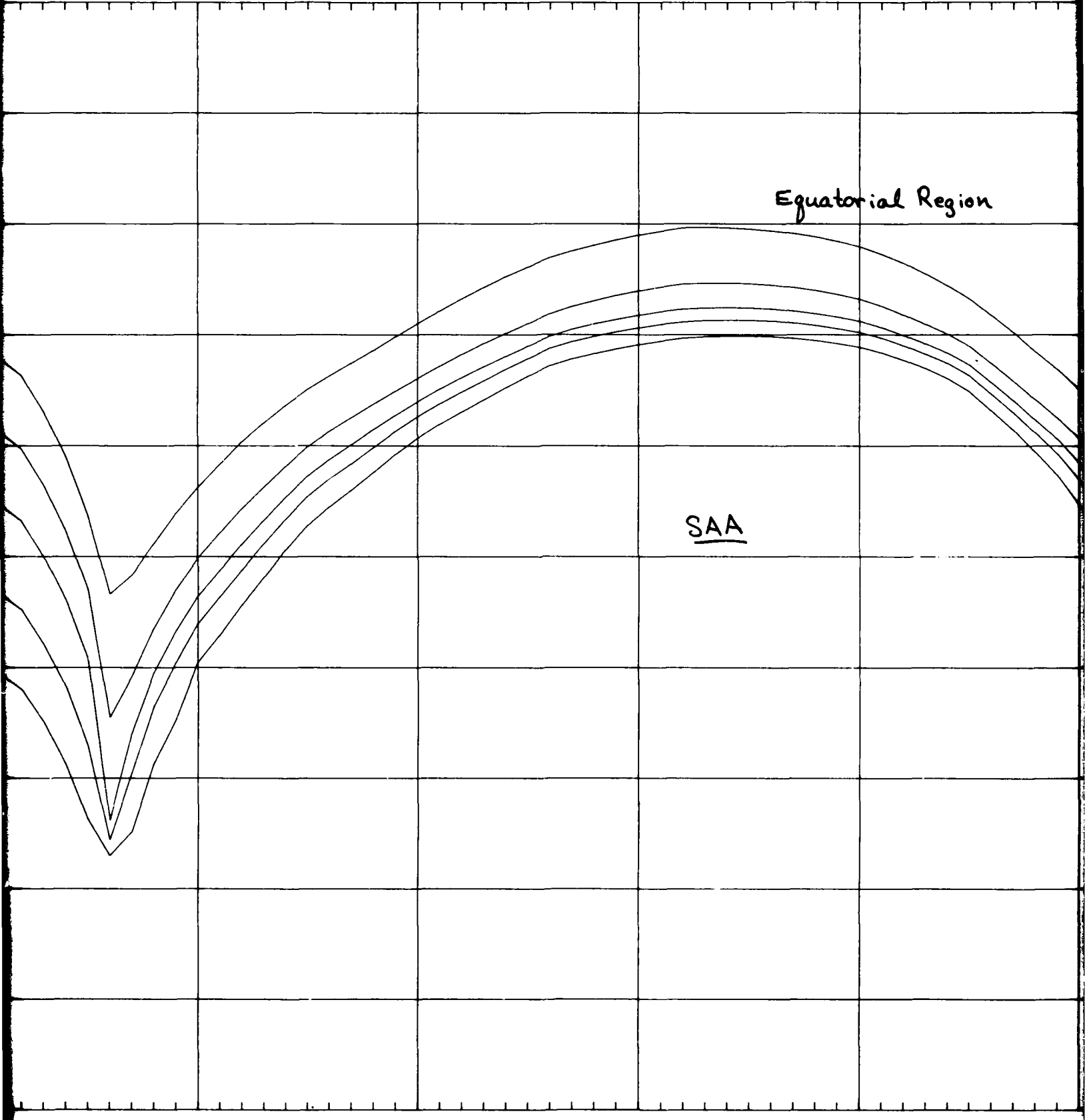
Southern Horn

SAA

40.0 50.0 60.0 70.0 80.0  
RELATIVE ORBIT TIME (MINUTES)

13

ITE ALUMINUM MEDIUM



Equatorial Region

SAA

TIME (MINUTES)

80.0

90.0

100.0

110.0

120.0

Figure 127

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

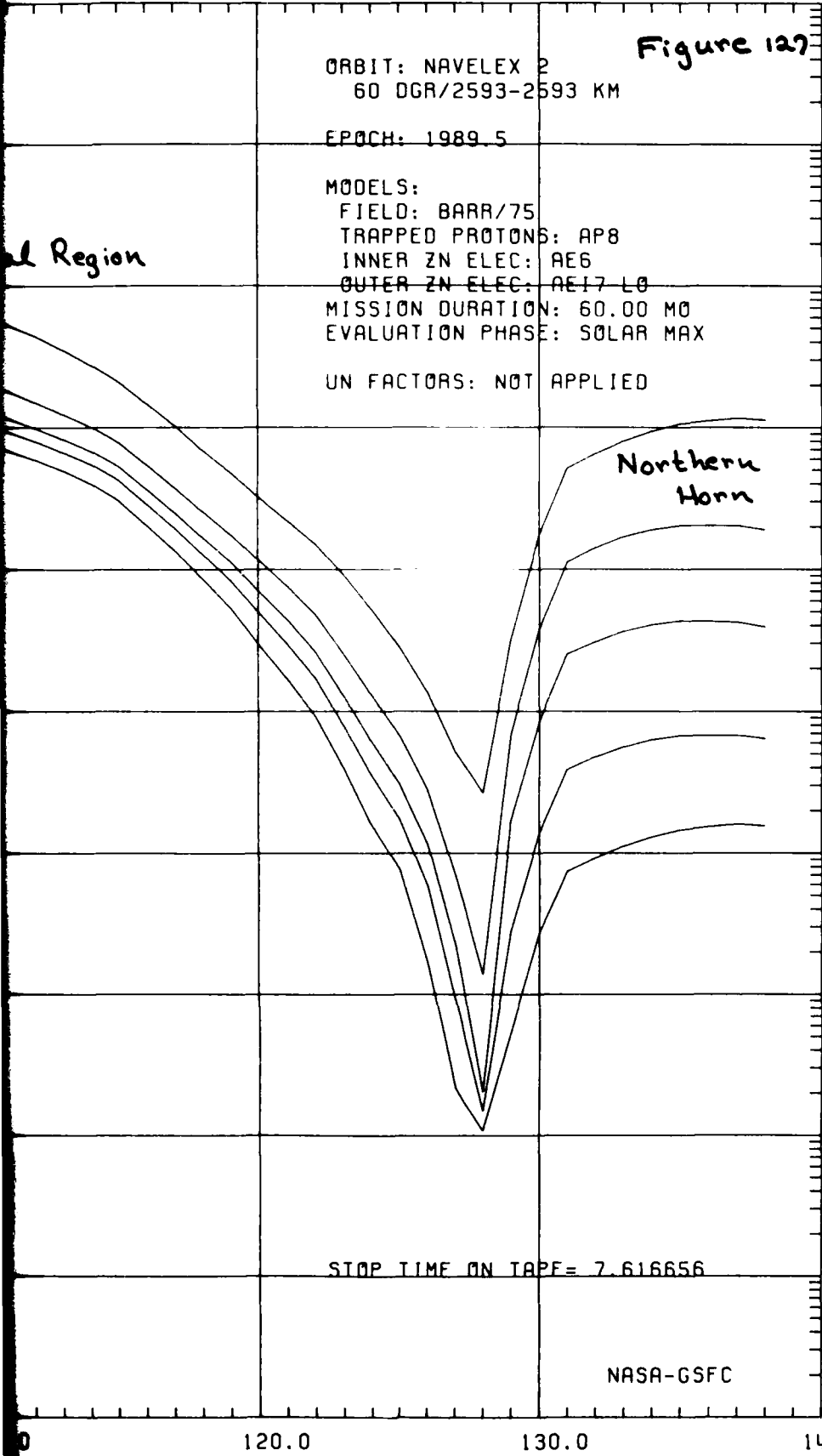
UN FACTORS: NOT APPLIED

al Region

Northern  
Horn

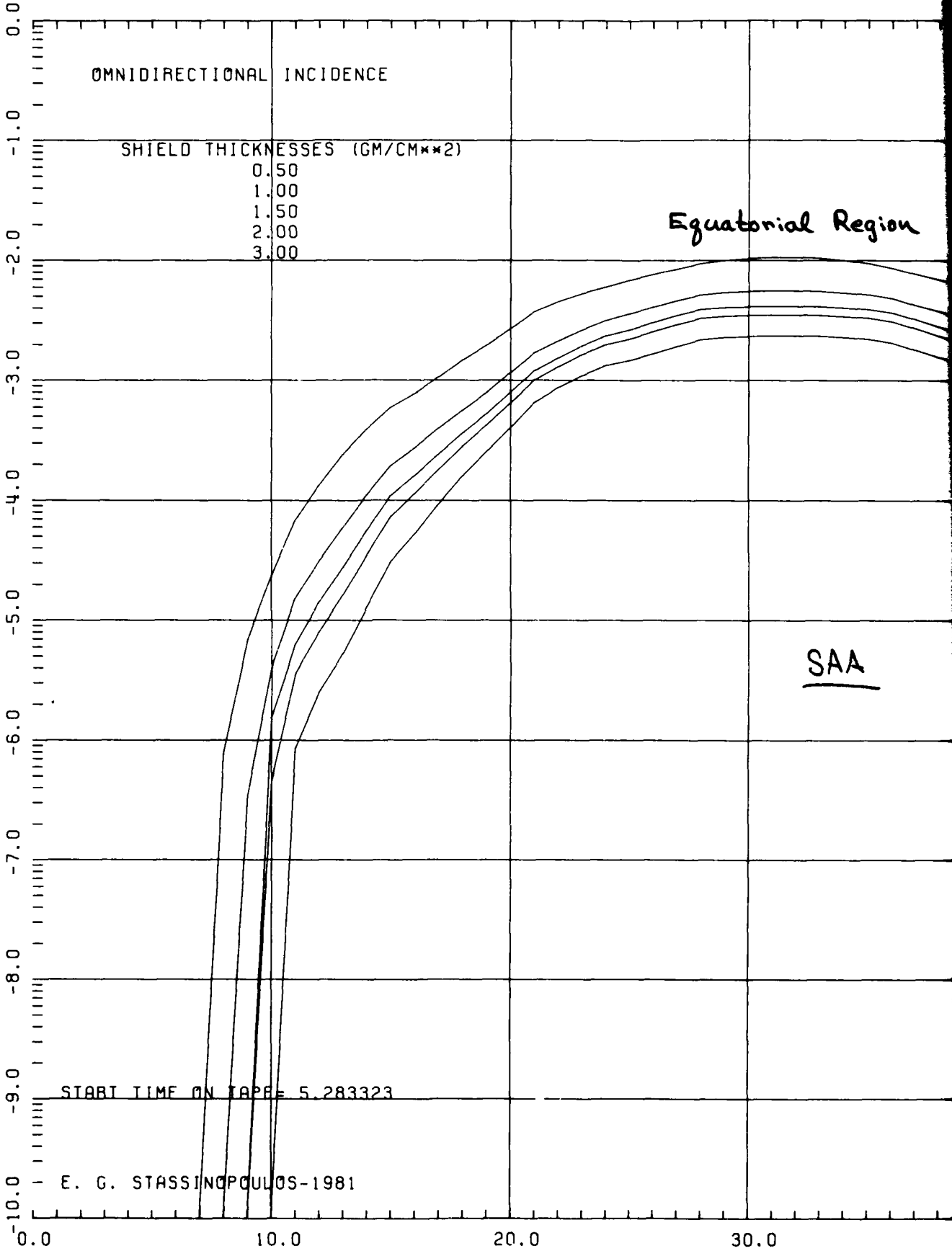
STOP TIME ON TAPE = 7.616656

NASA-GSFC



INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)

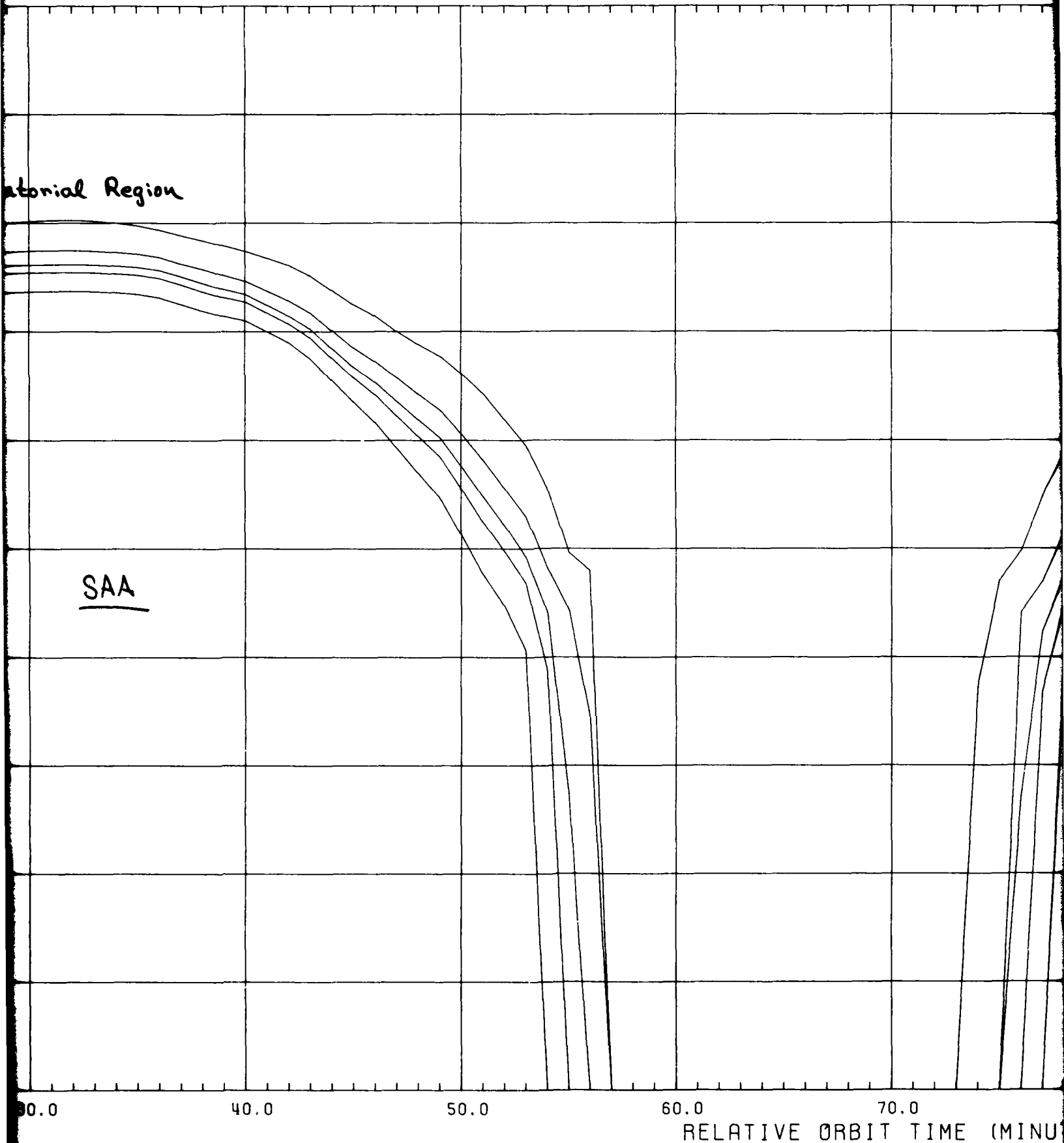




DOSE AT CENTER OF ALUMINUM

atorial Region

SAA



RELATIVE ORBIT TIME (MINU)

OF ALUMINUM SPHERES

Equatorial Region

SAA

80.0 90.0 100.0 110.0  
TIME (MINUTES)

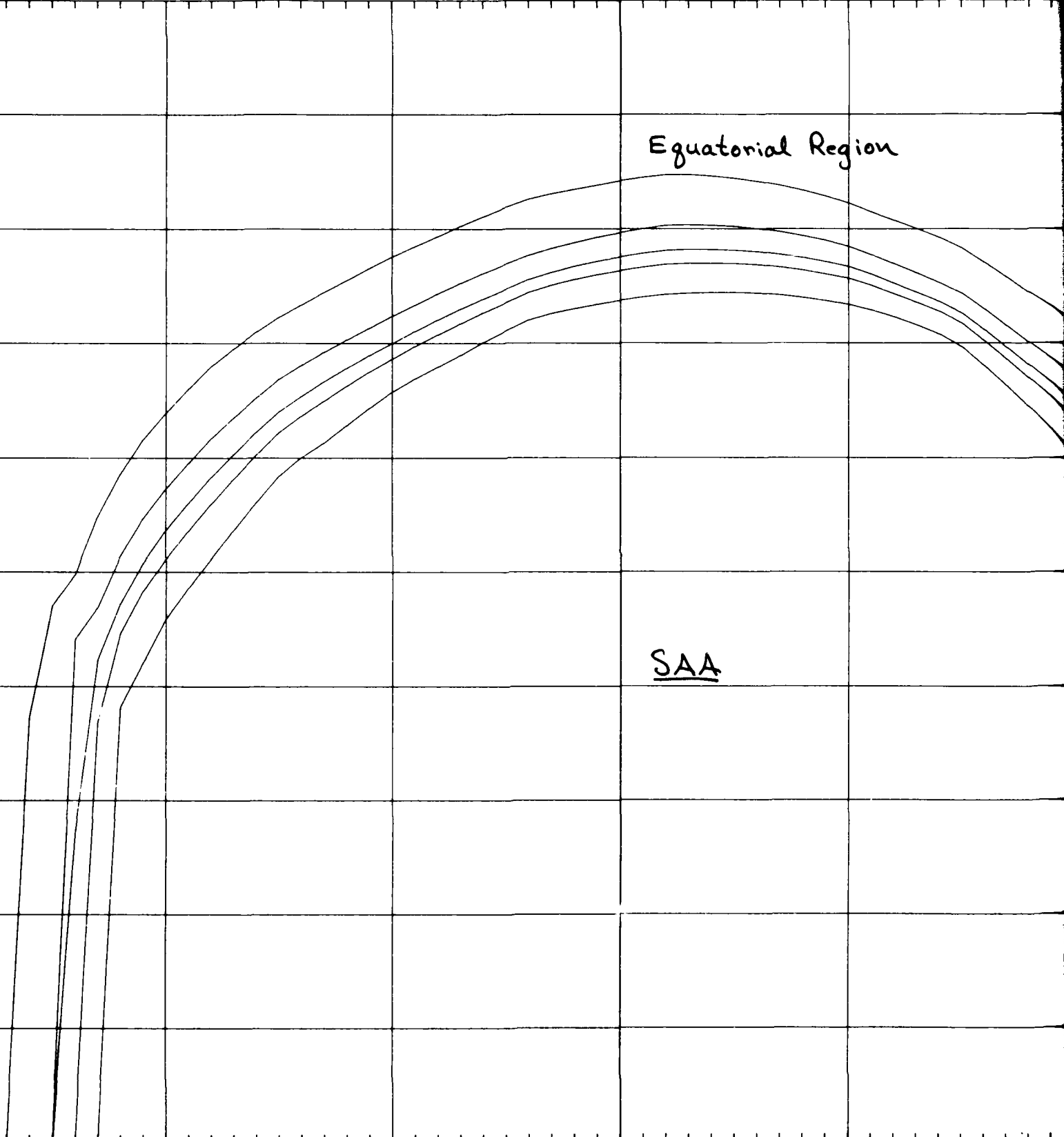


Figure 128

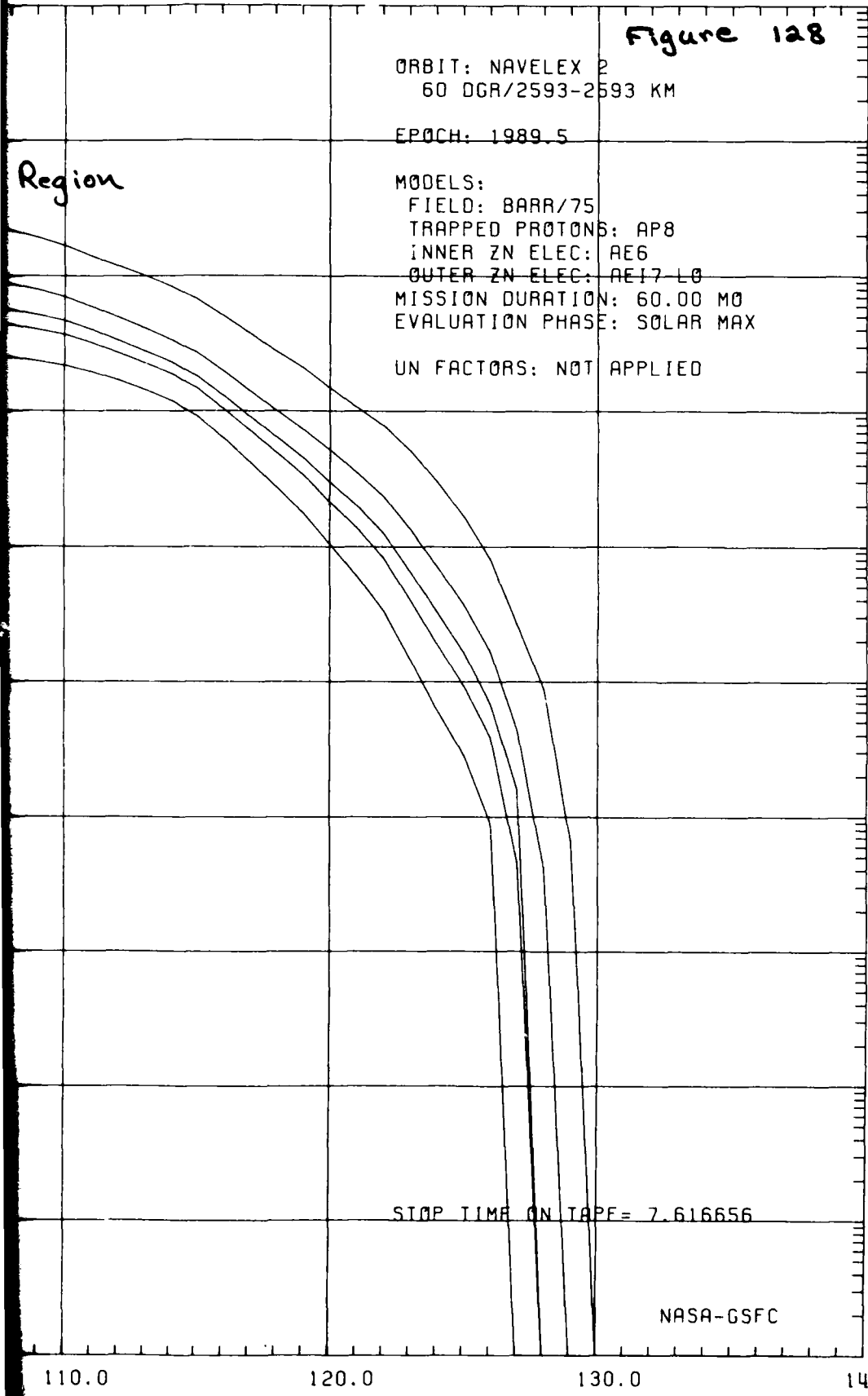
ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Region



STOP TIME ON TAPE = 7.616656

NASA-GSFC

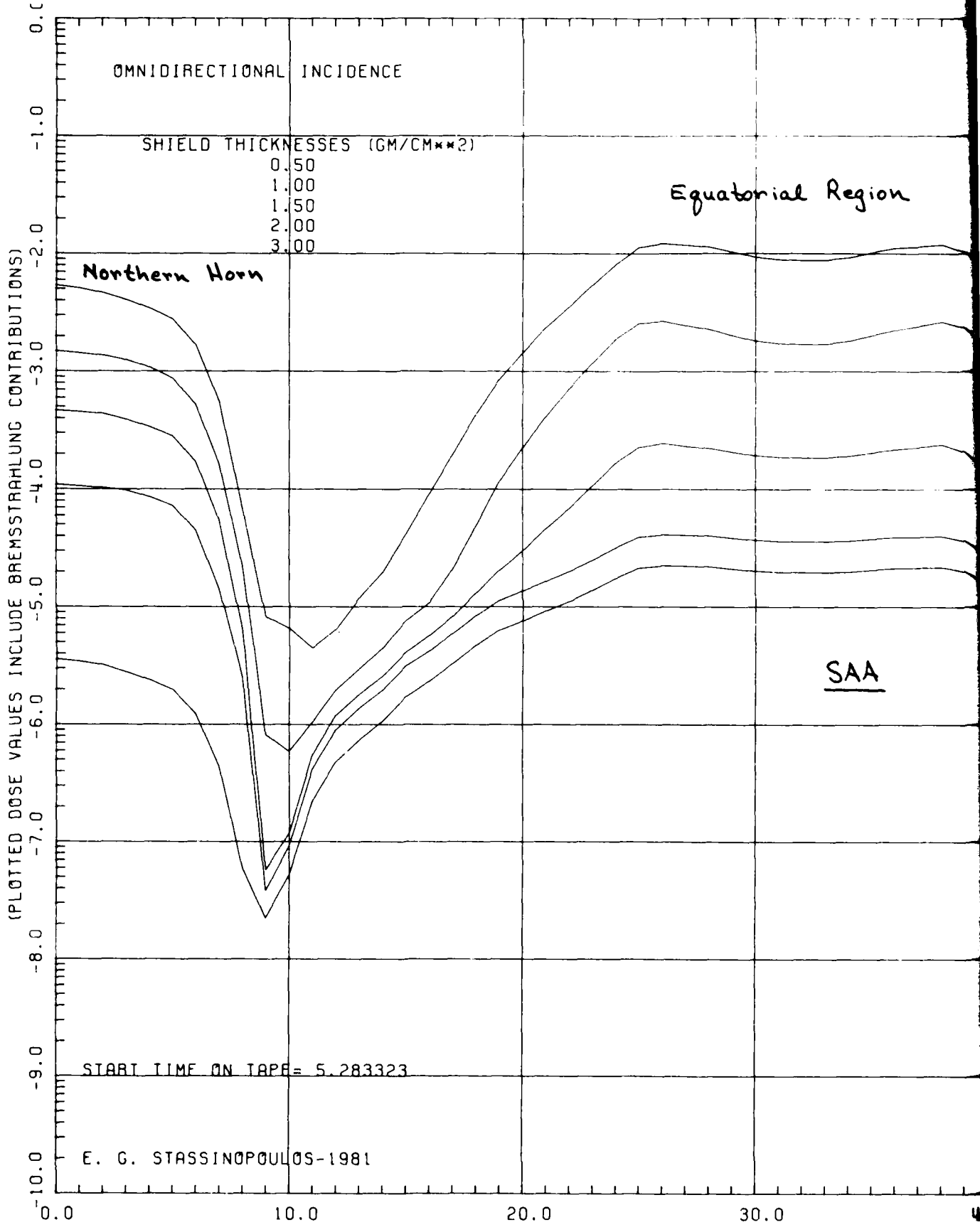
110.0

120.0

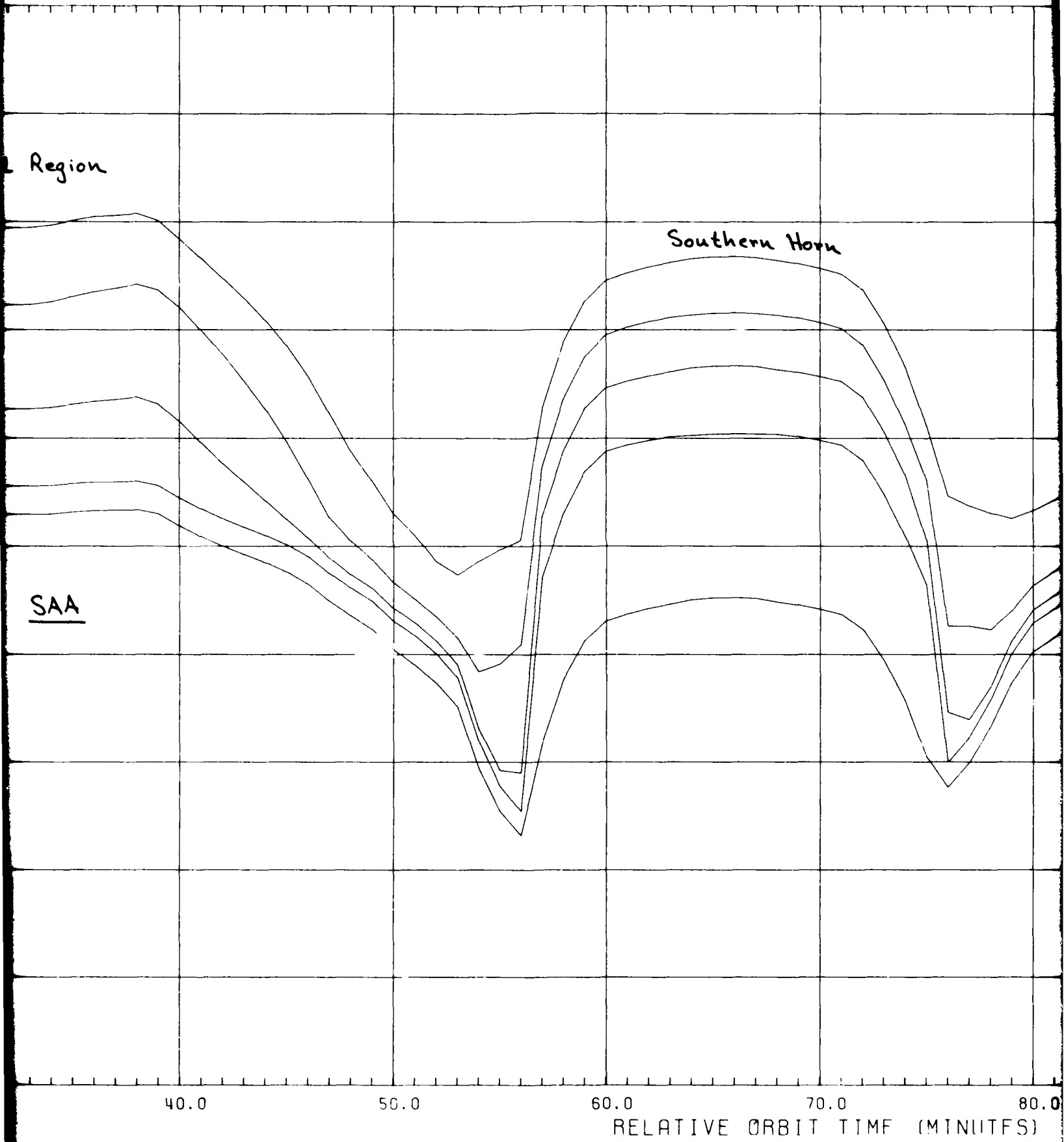
130.0

140.0

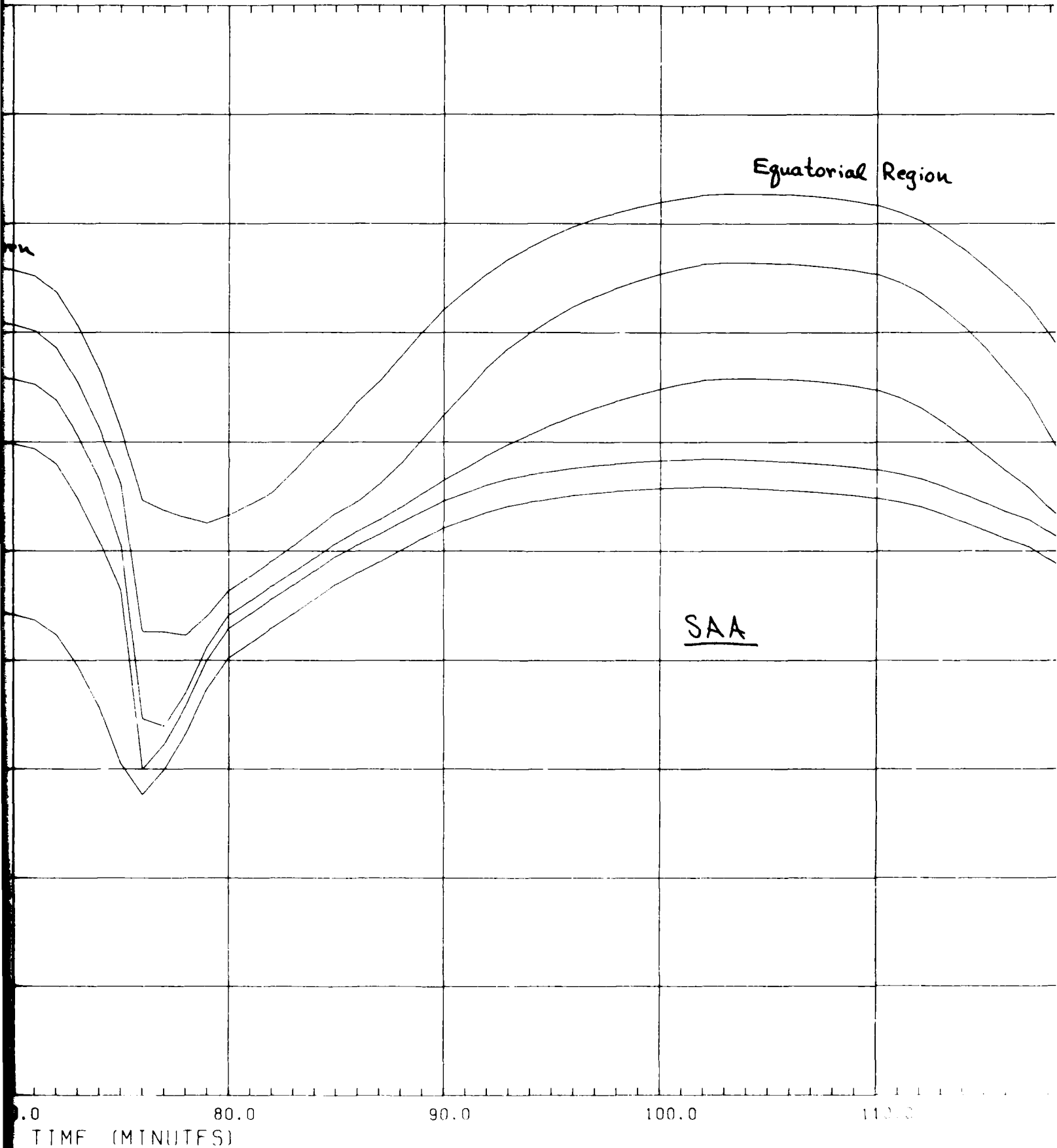
INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)



DOSE AT CENTER OF ALUMINUM SPH



OF ALUMINUM SPHERES



Equatorial Region

SAA

0.0 80.0 90.0 100.0 110.0  
TIME (MINUTES)

AD-A141 849

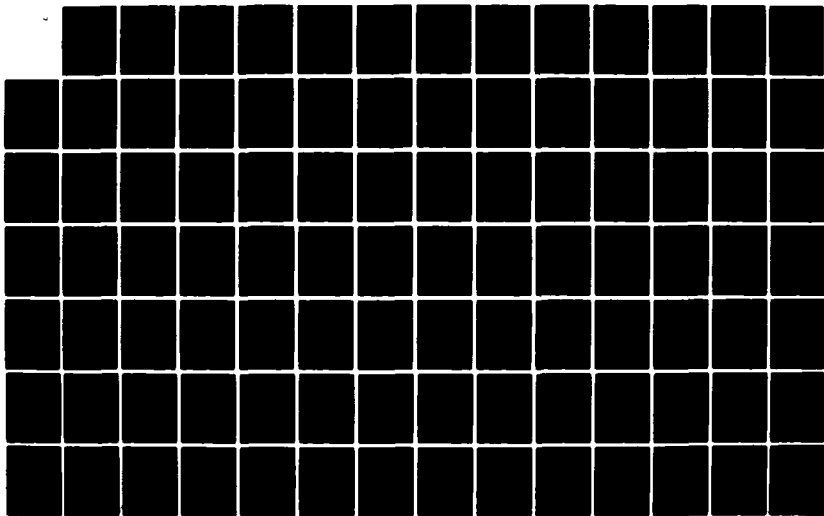
ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR  
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE  
ADMINISTRATION GREENBELT MD GO. . E G STASSINOPOULOS  
NOV 81 NASA-GSFC-X-601-81-28

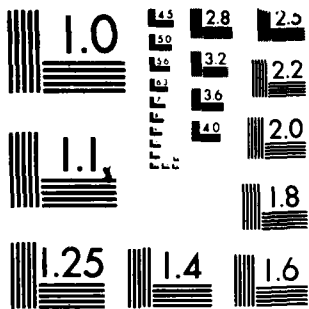
4/5

UNCLASSIFIED

F/G 22/3

NL





MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS 1963-A



4

Figure 129

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

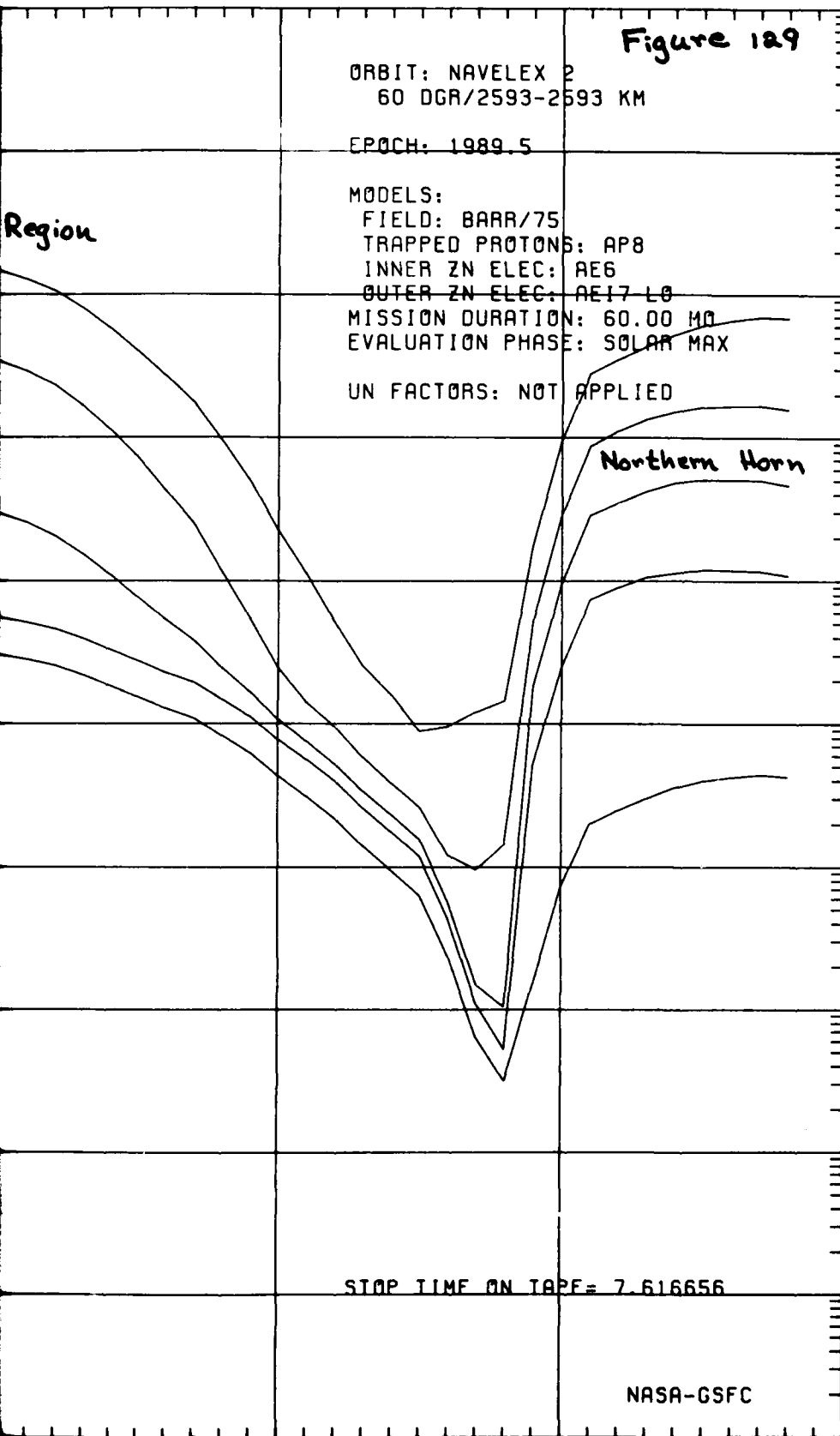
Region

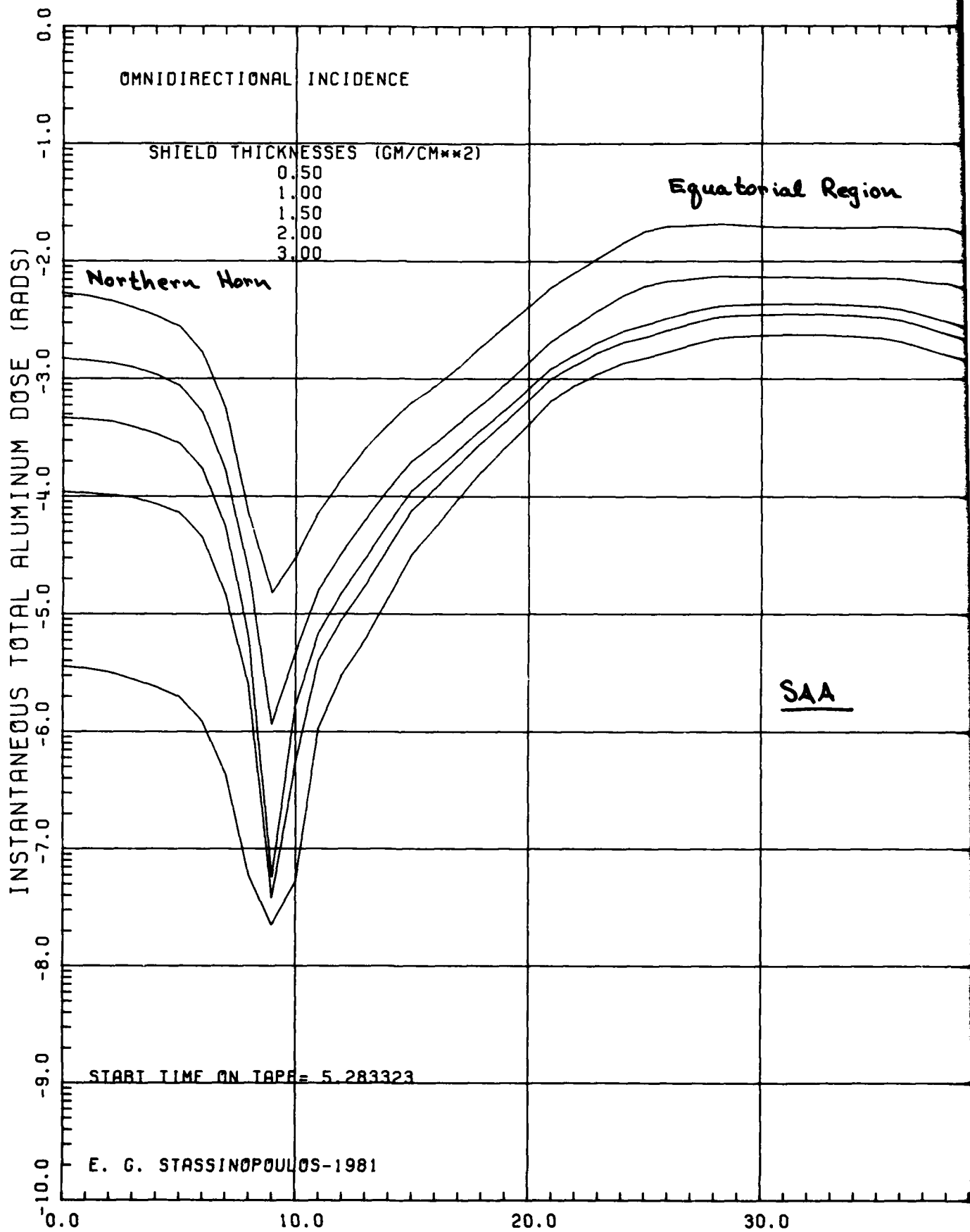
Northern Horn

STOP TIME ON TAPE = 7.616656

NASA-GSFC

0.0 120.0 130.0 140.0





1  
2

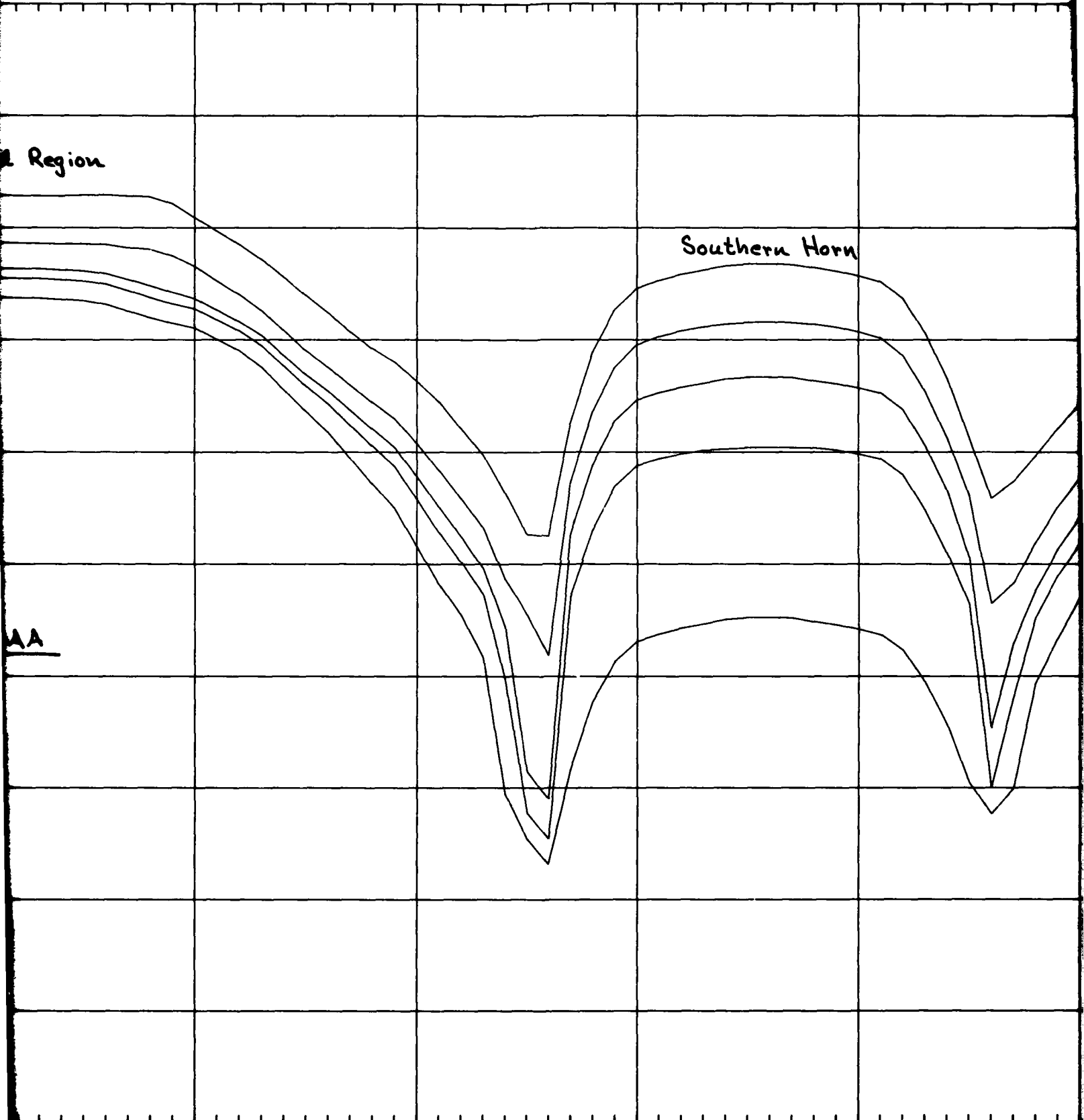
DOSE AT CENTER OF ALUMINUM S

Region

Southern Horn

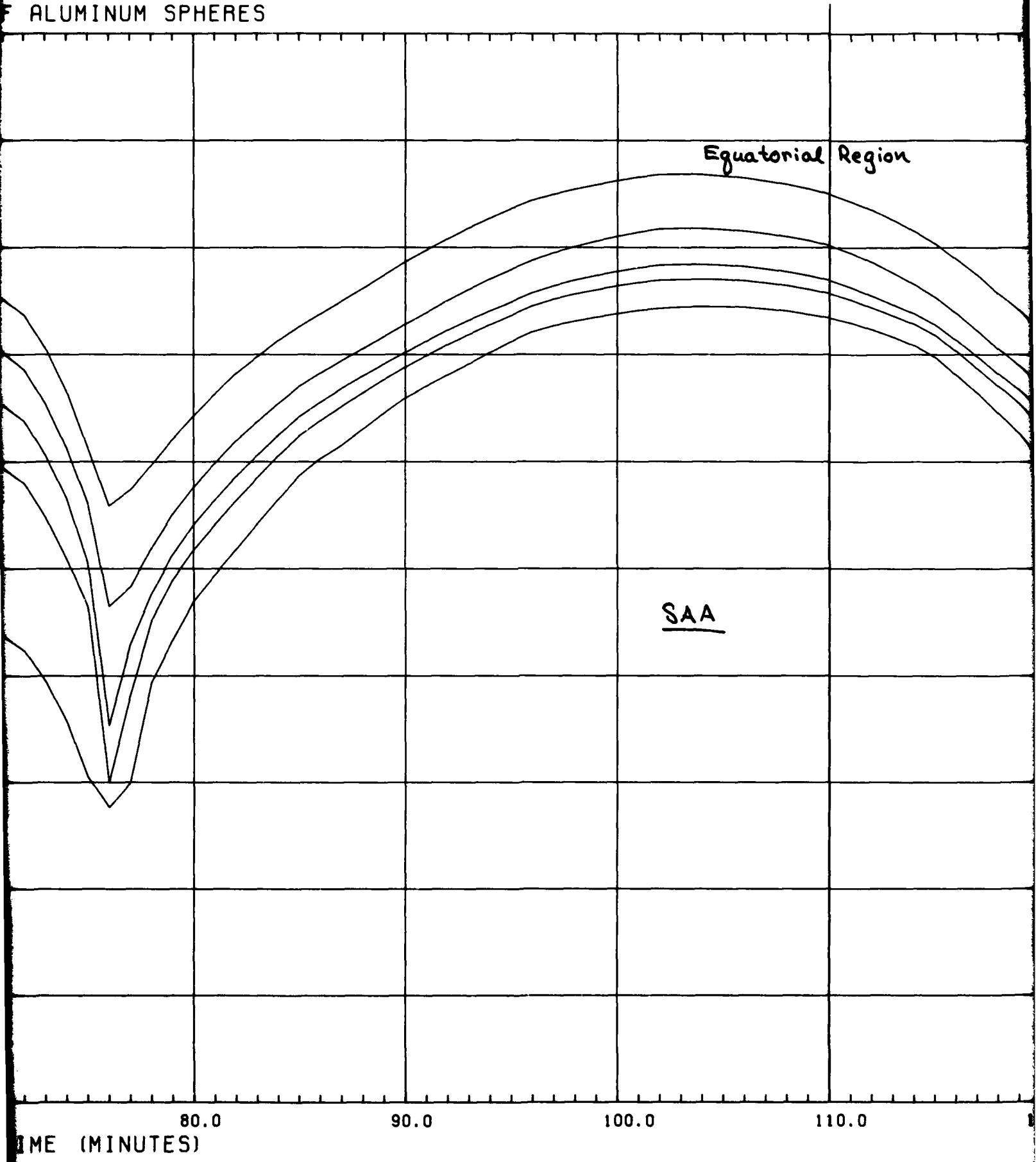
AA

40.0 50.0 60.0 70.0 80  
RELATIVE ORBIT TIME (MINUTES)



13

ALUMINUM SPHERES



4

Figure 130

ORBIT: NAVELEX 2  
60 DGR/2593-2593 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

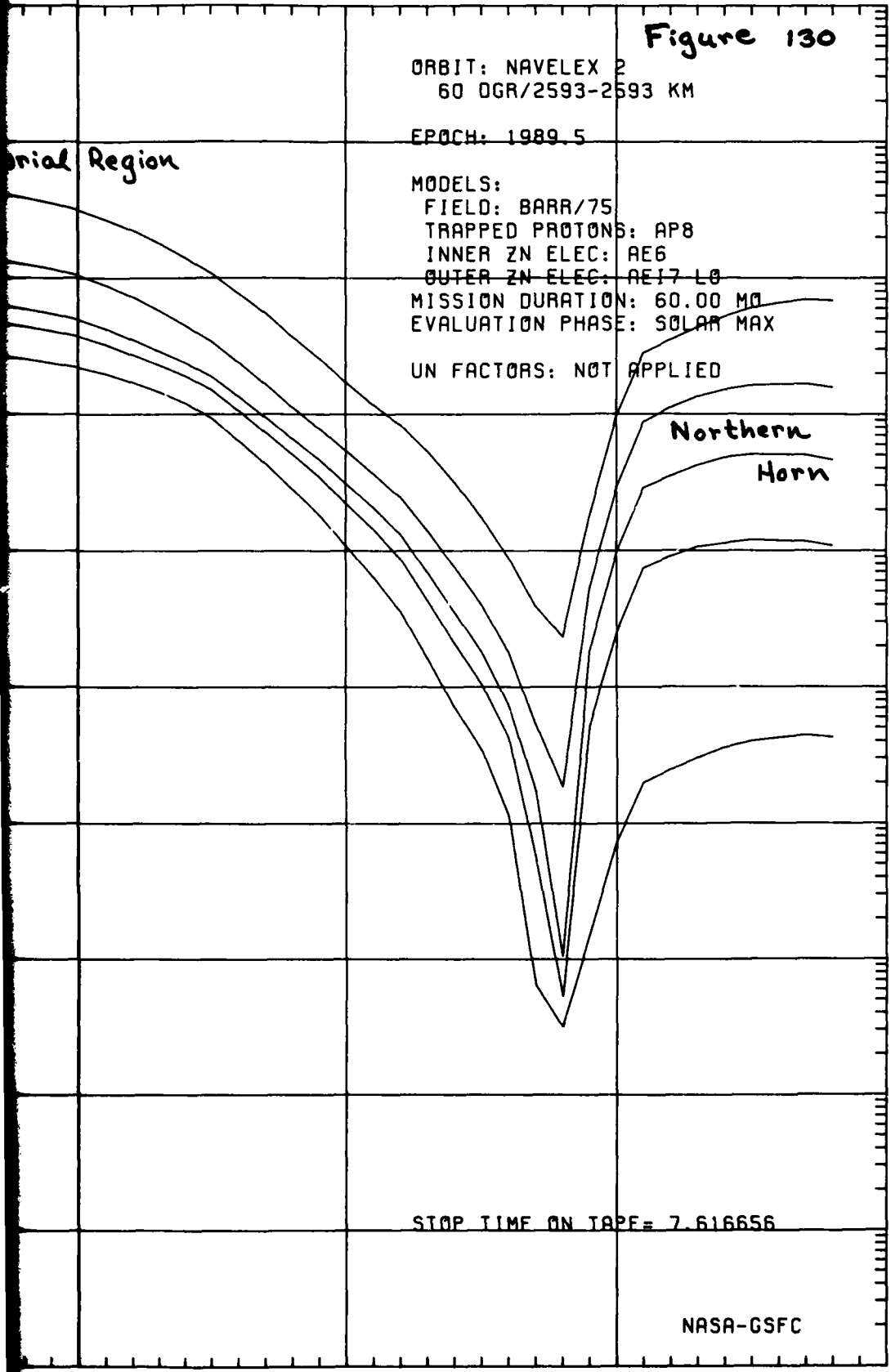
Orbital Region

Northern  
Horn

STOP TIME ON TAPE = 7.616656

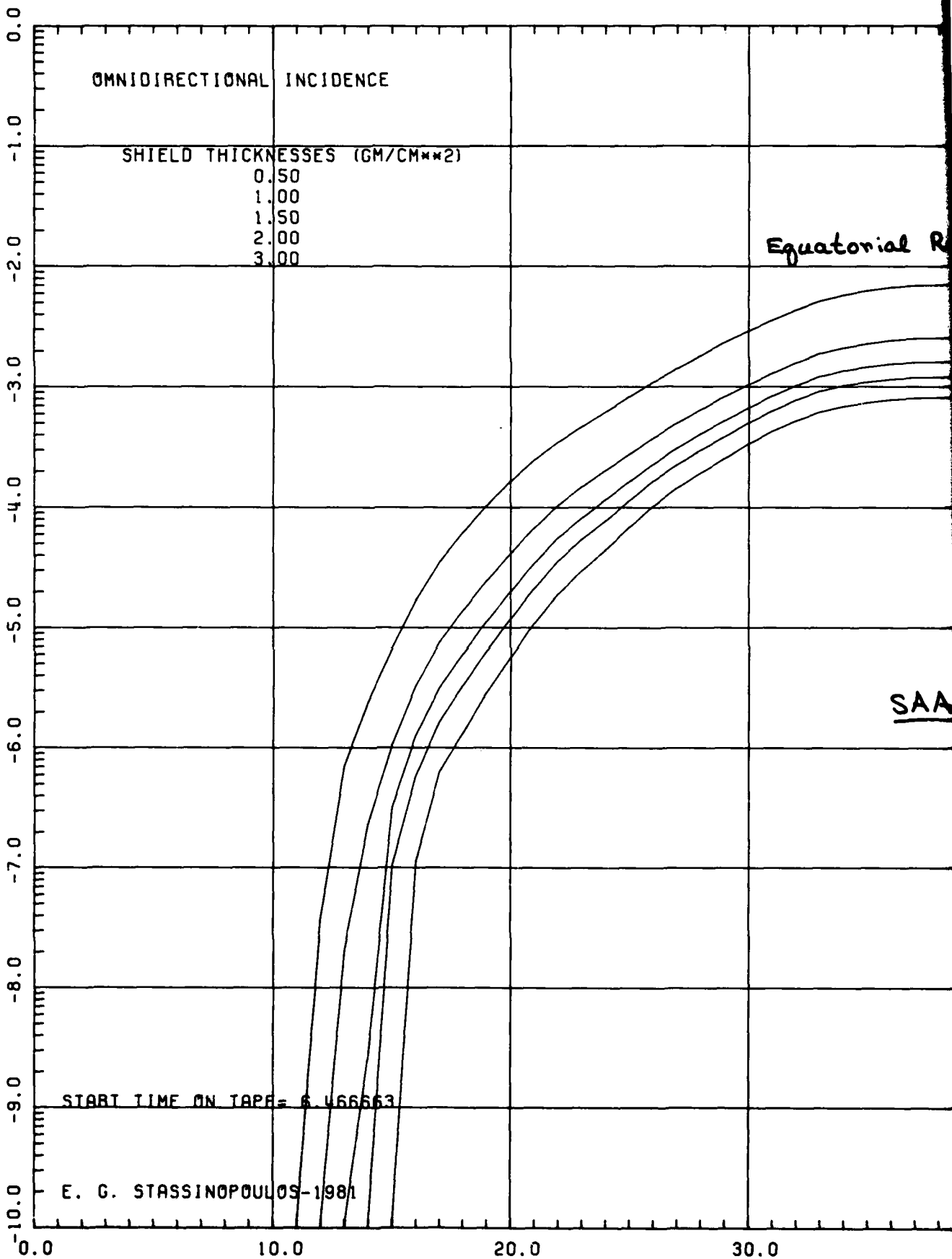
NASA-GSFC

110.0 120.0 130.0 140.0



INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



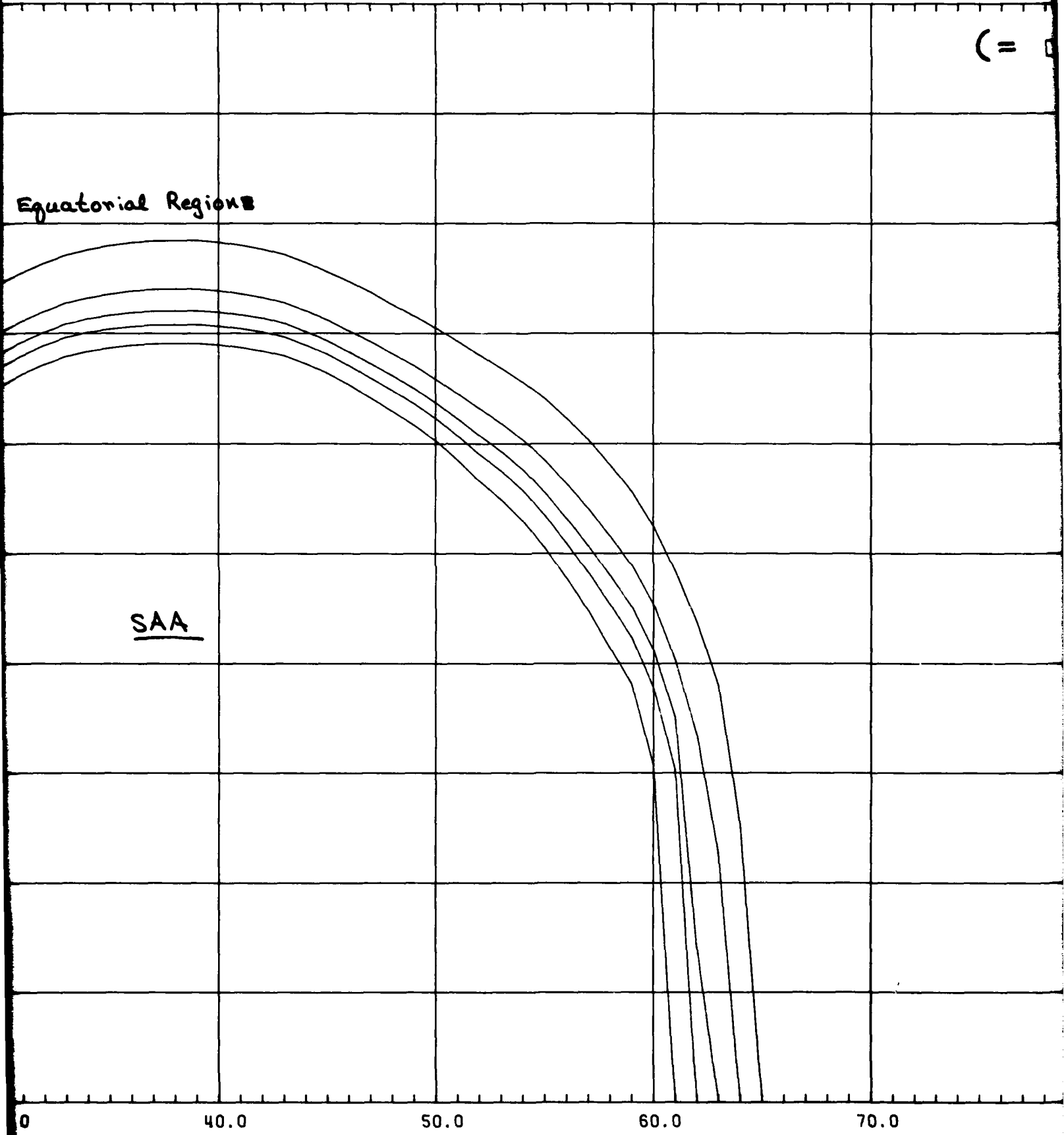
2

DOSE AT T

( =

Equatorial Regions

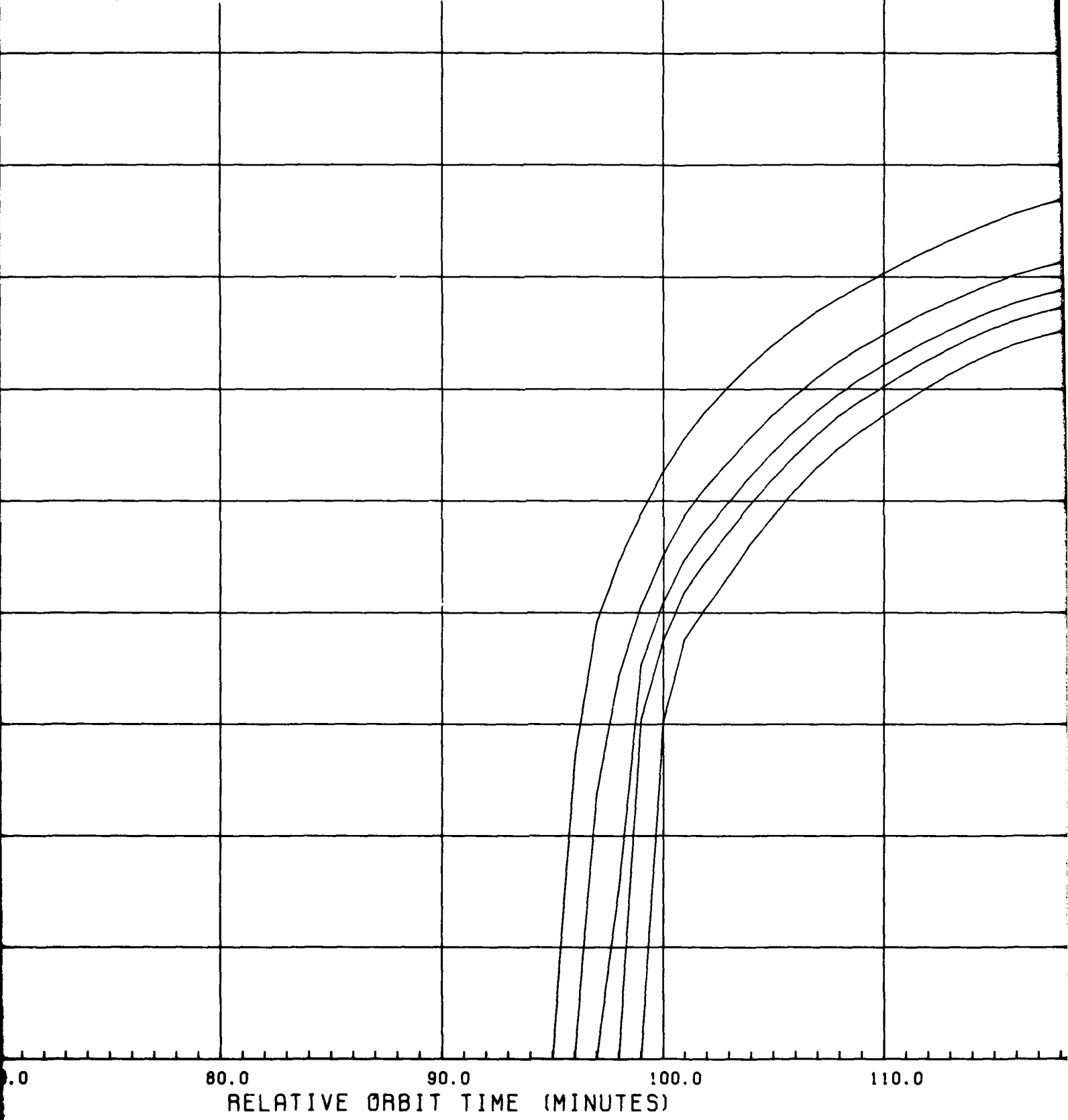
SAA



13

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS

( = DOSE IN SEMI-INFINITE ALUMINUM MEDIUM )





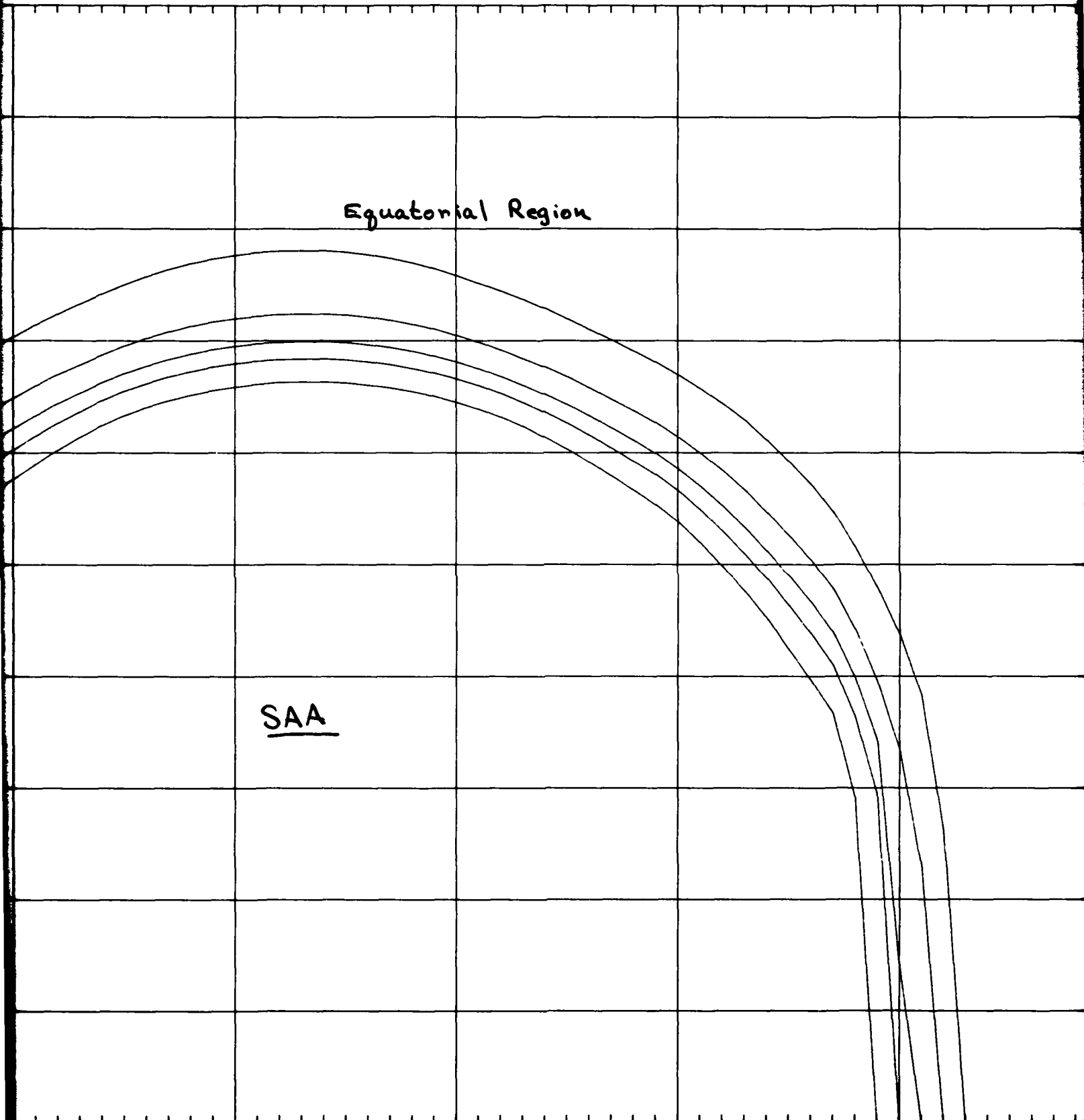
14

HIELDS

Equatorial Region

SAA

0.0 120.0 130.0 140.0 150.0



5

Figure 131

ORBIT: NAVELEX B  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 9.316662

NASA-GSFC

150.0

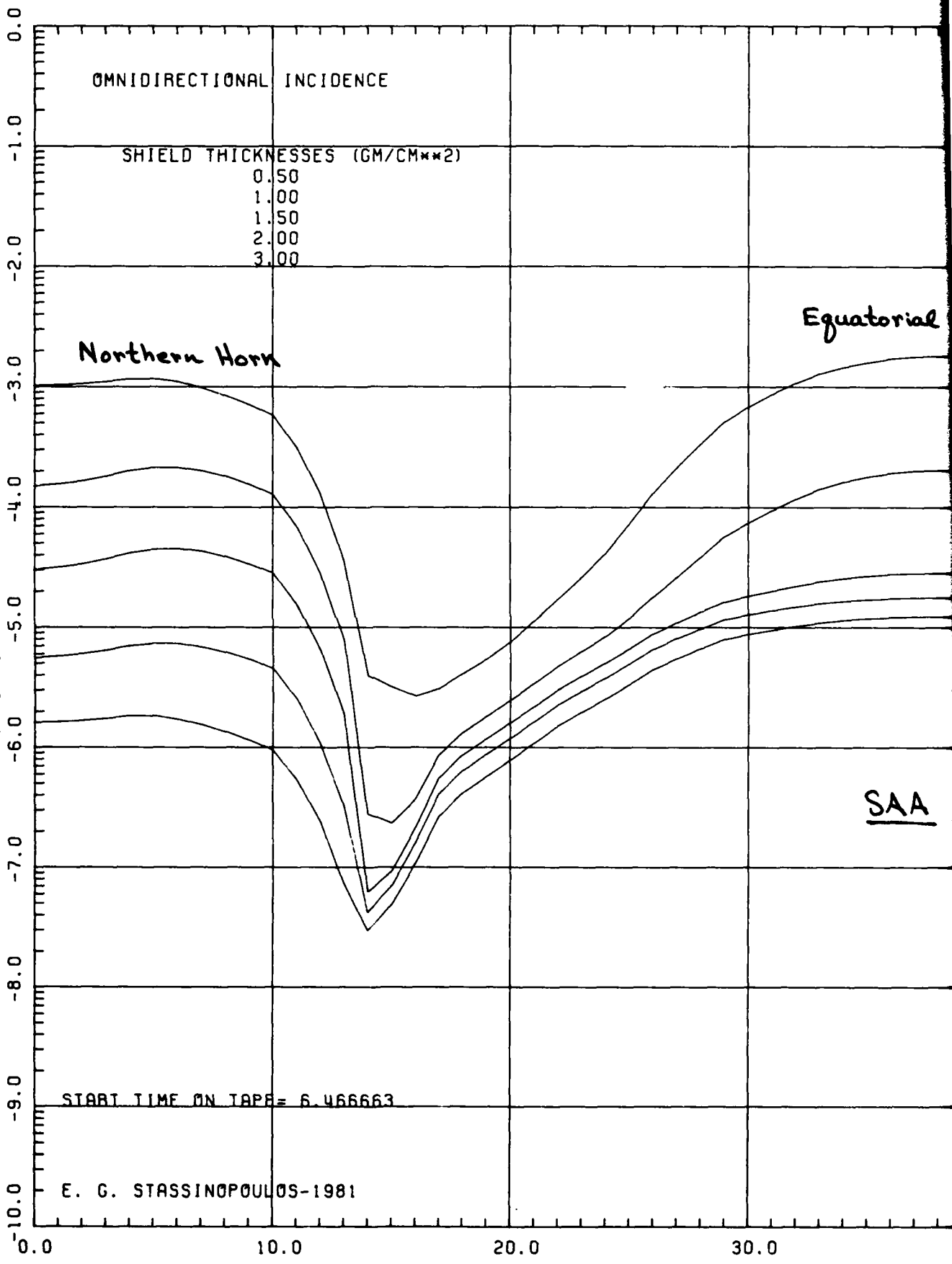
160.0

170.0

180.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



1 2

DOSE AT TRA

Equatorial Region

Southern Horn

SAA

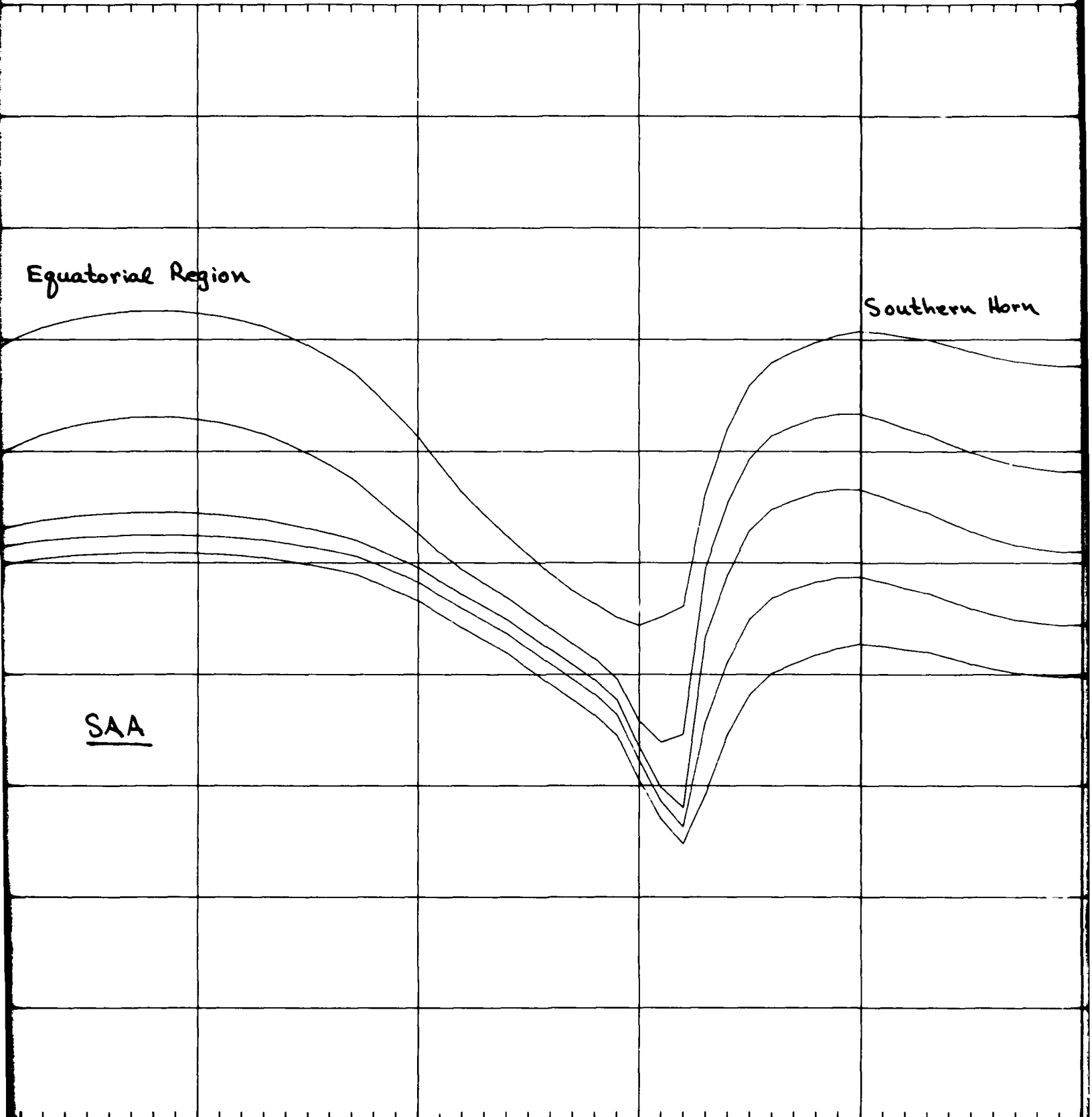
40.0

50.0

60.0

70.0

80.0

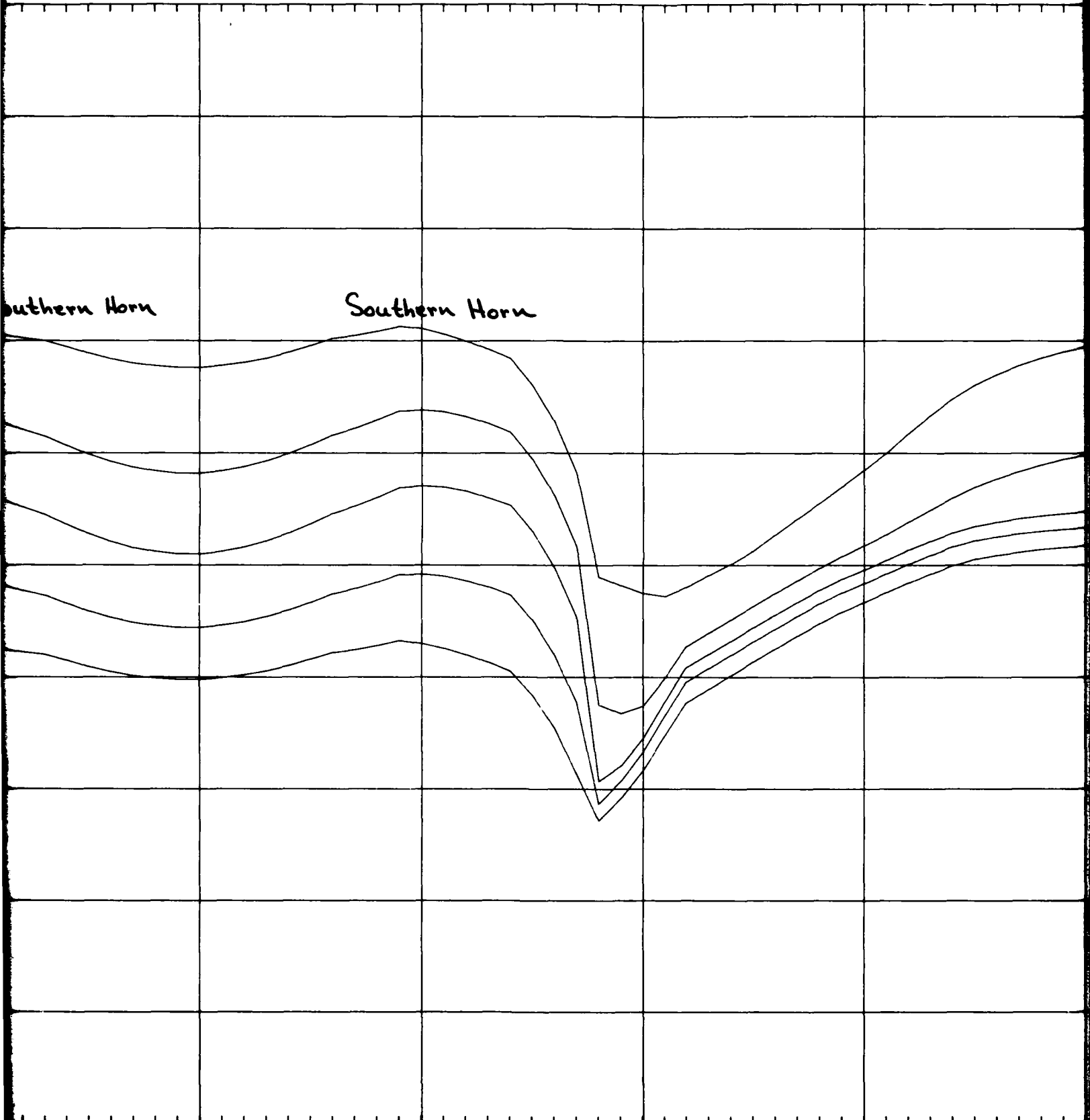


13

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS

Southern Horn

Southern Horn



80.0

90.0

100.0

110.0

RELATIVE ORBIT TIME (MINUTES)

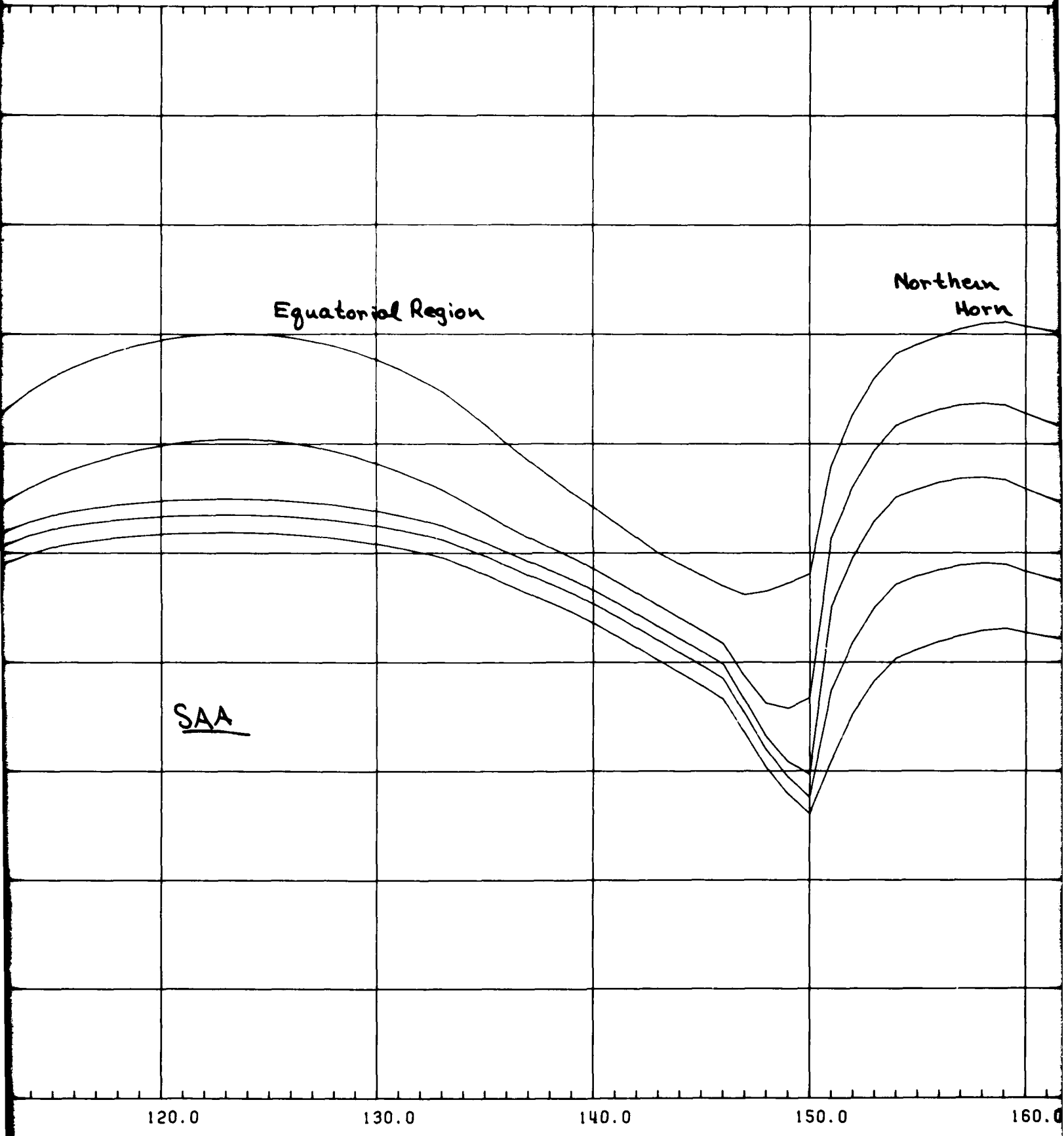


Figure 132

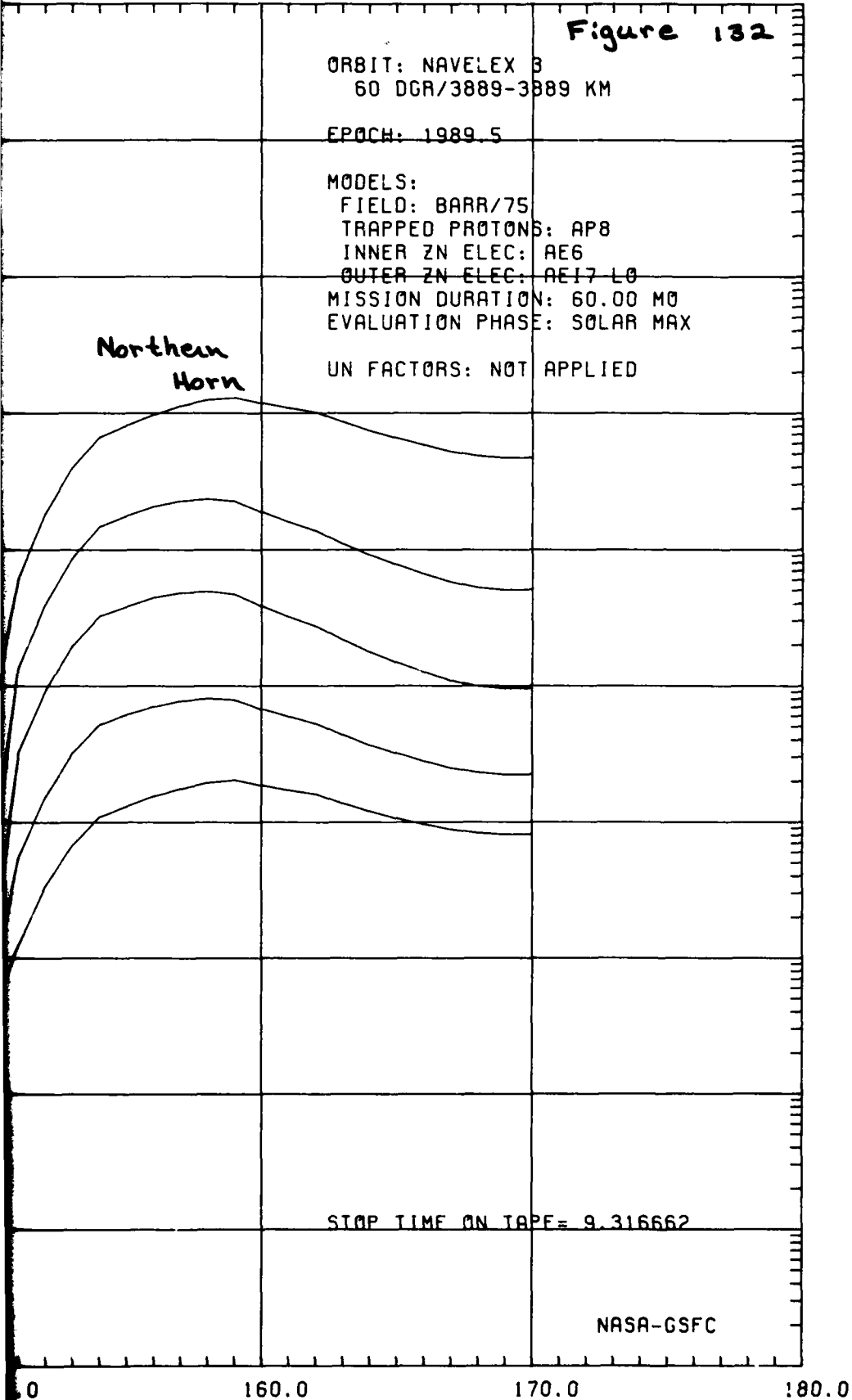
ORBIT: NAVELEX B  
60 DGR/3889-3889 KM

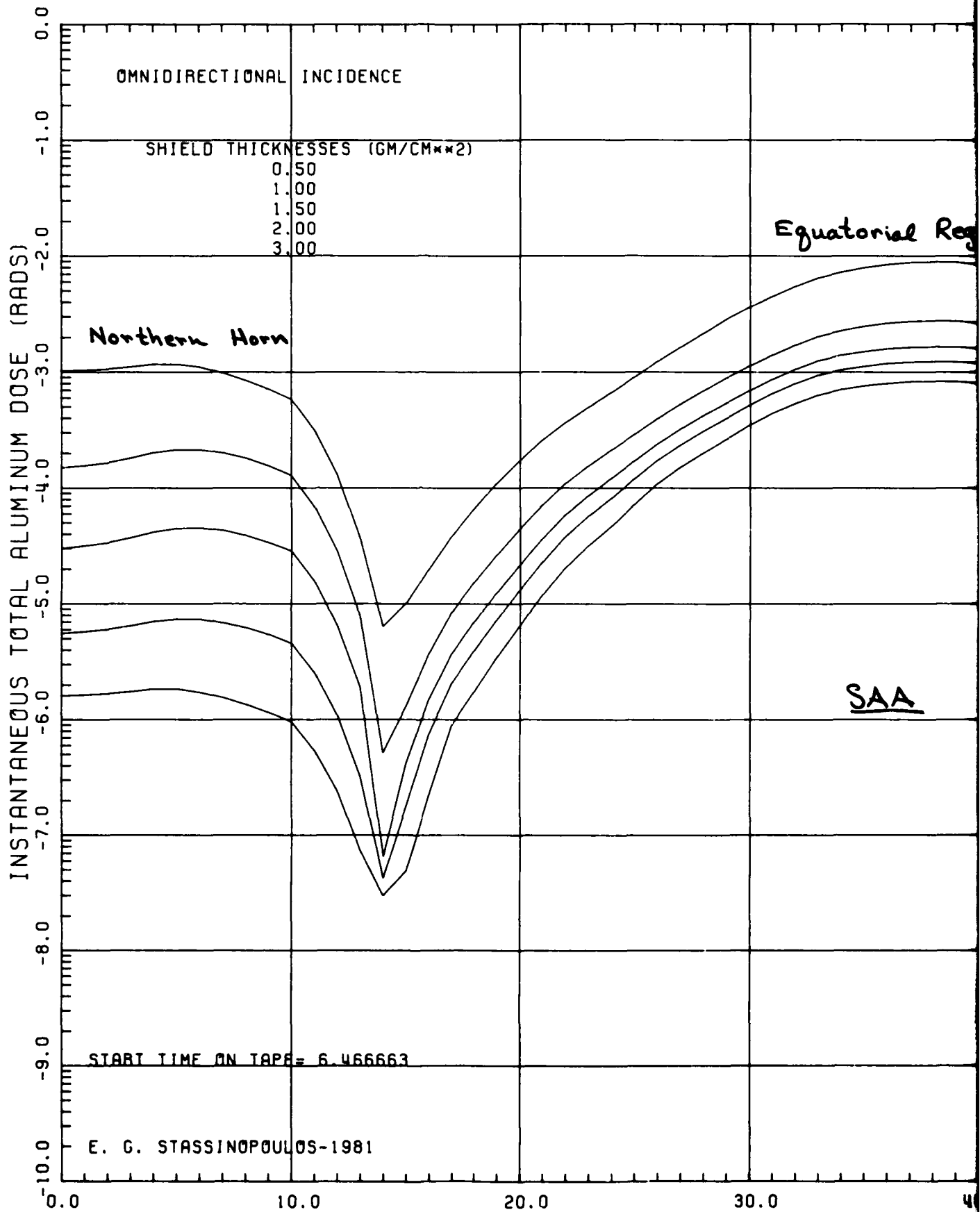
EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn







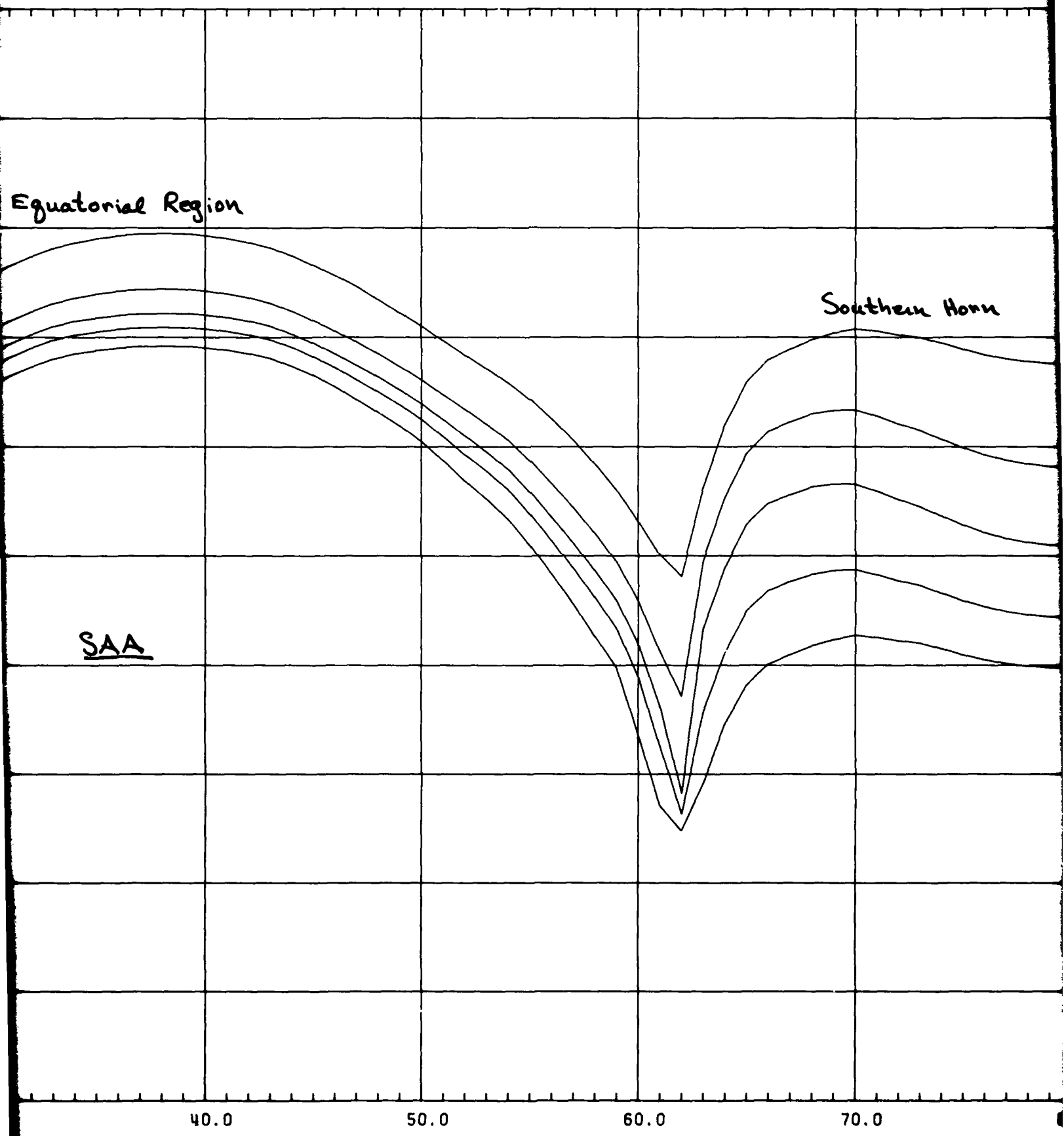
2

DOSE AT TR

Equatorial Region

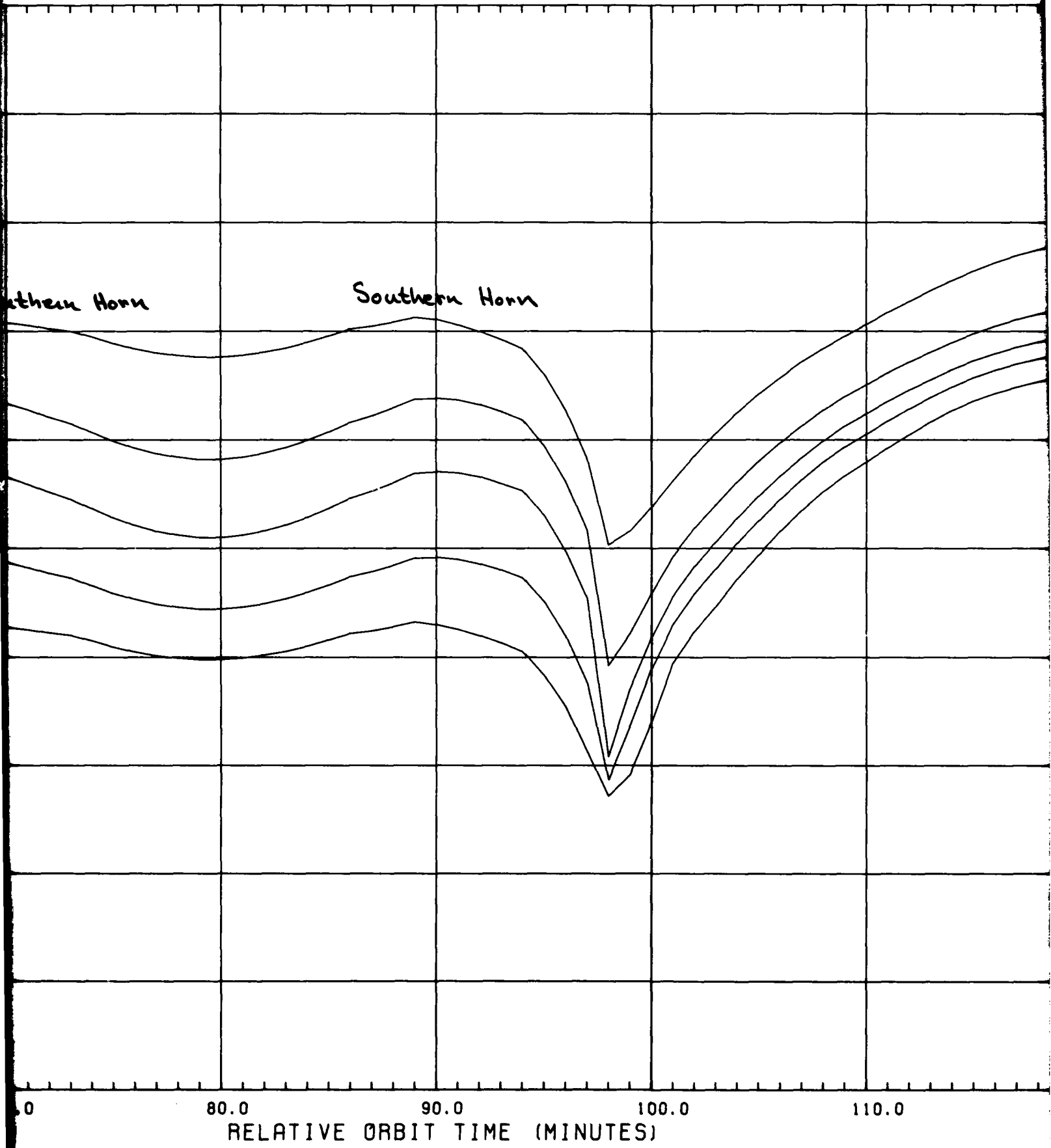
Southern Horn

SAA



13

# DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM SLAB SHIELDS



4

FIELDS

Equatorial Region

Northern  
Horn

SAA

0.0 120.0 130.0 140.0 150.0

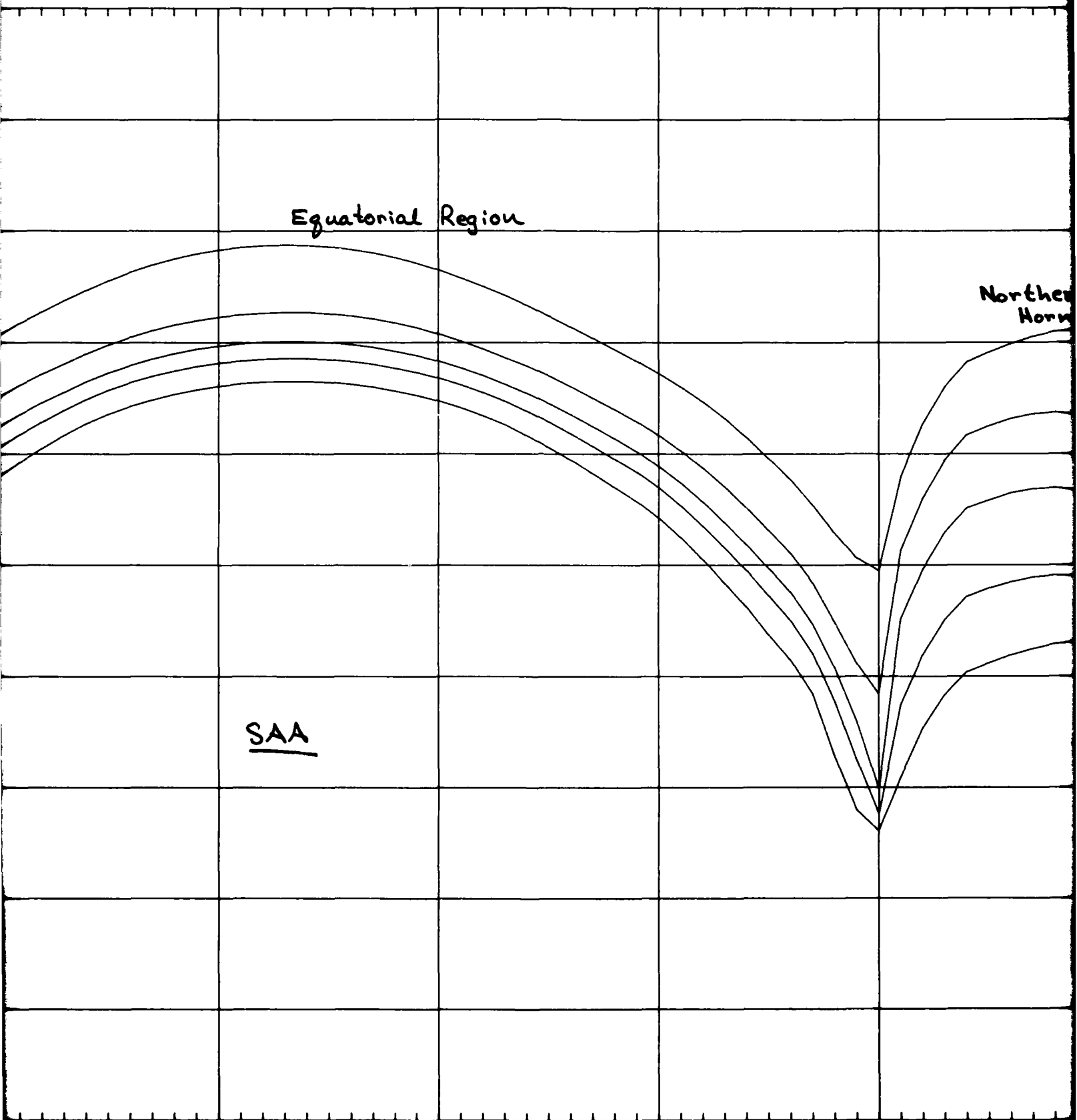


Figure 133

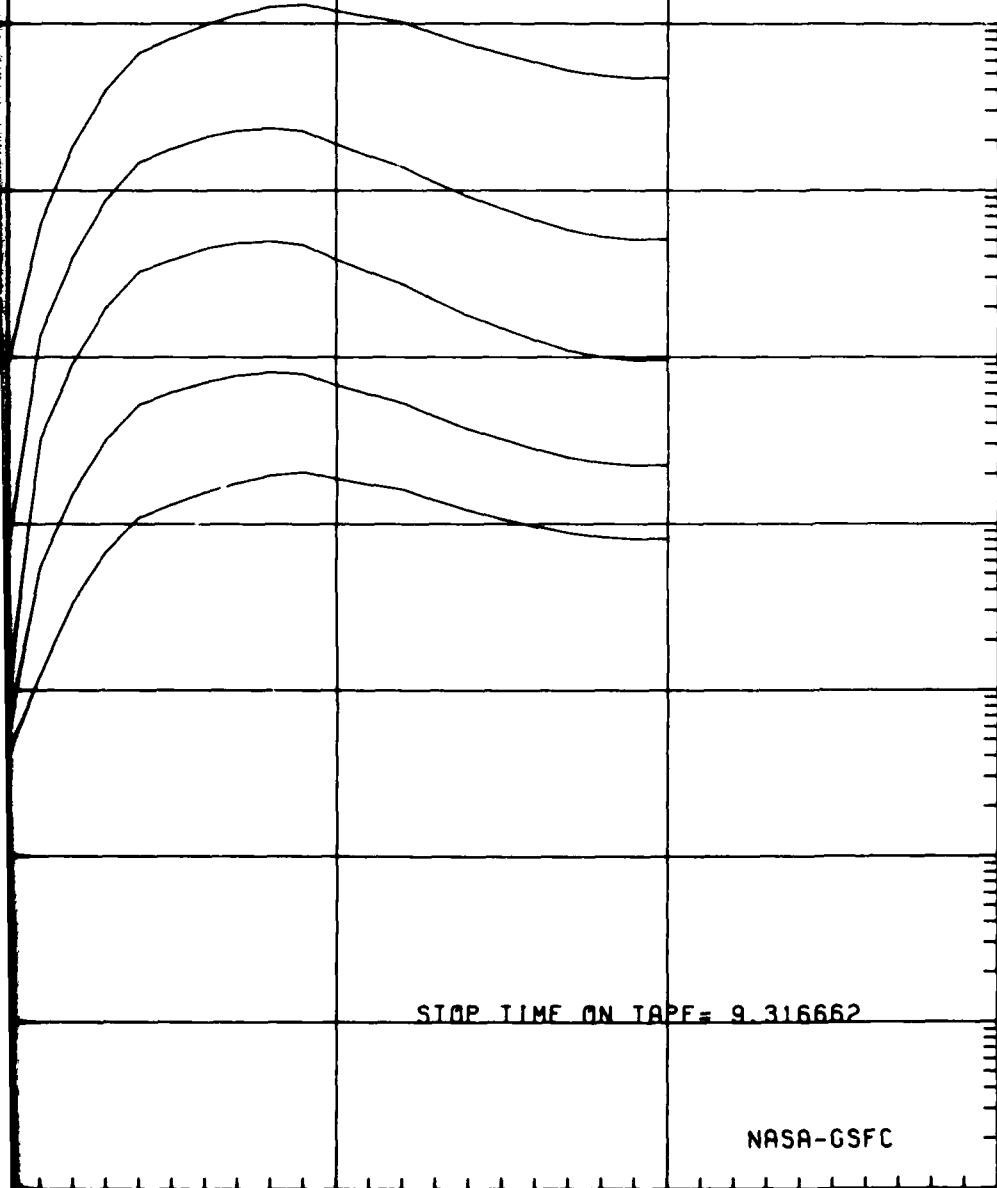
ORBIT: NAVELEX 3  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn



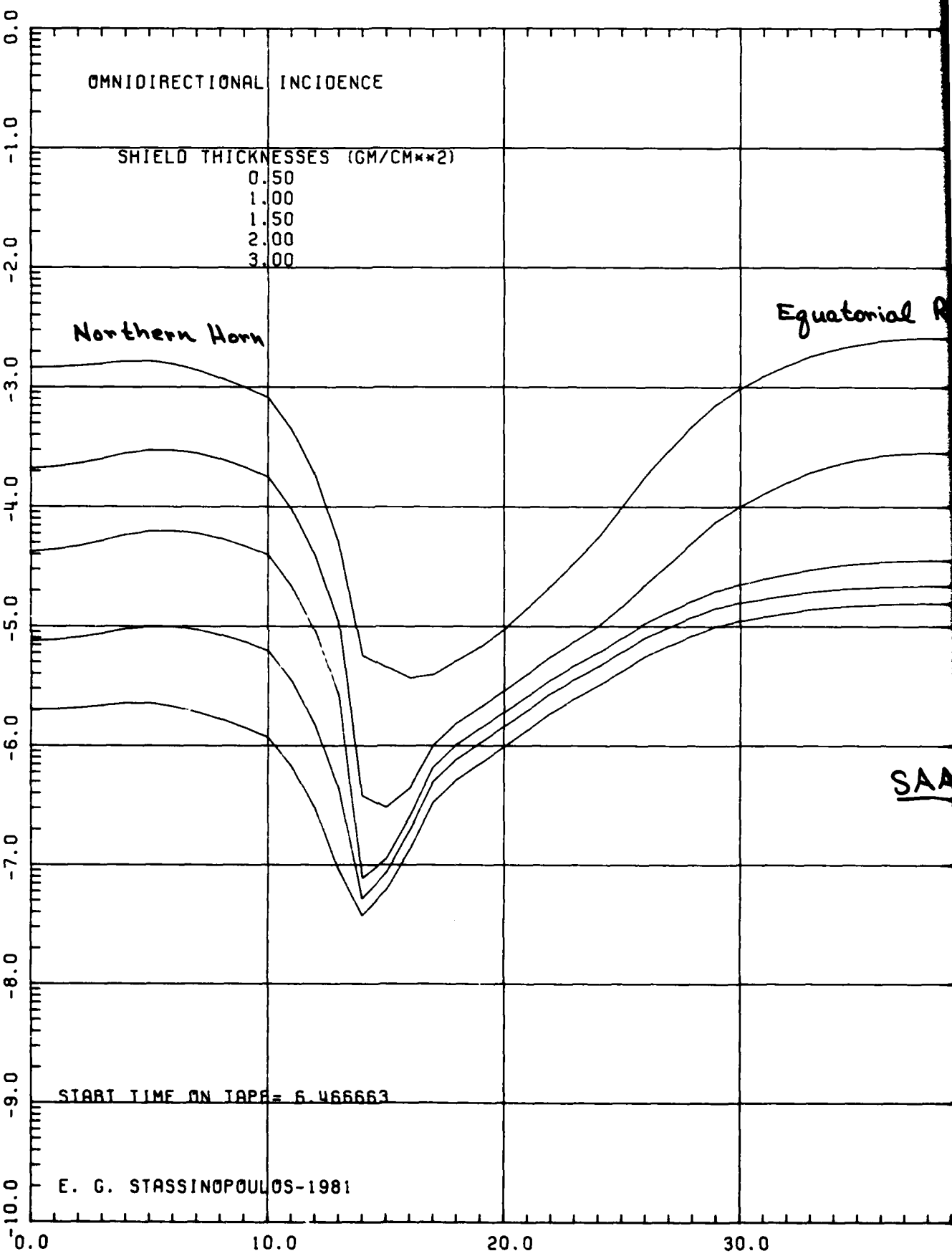
STOP TIME ON TAPE = 9.316662

NASA-GSFC

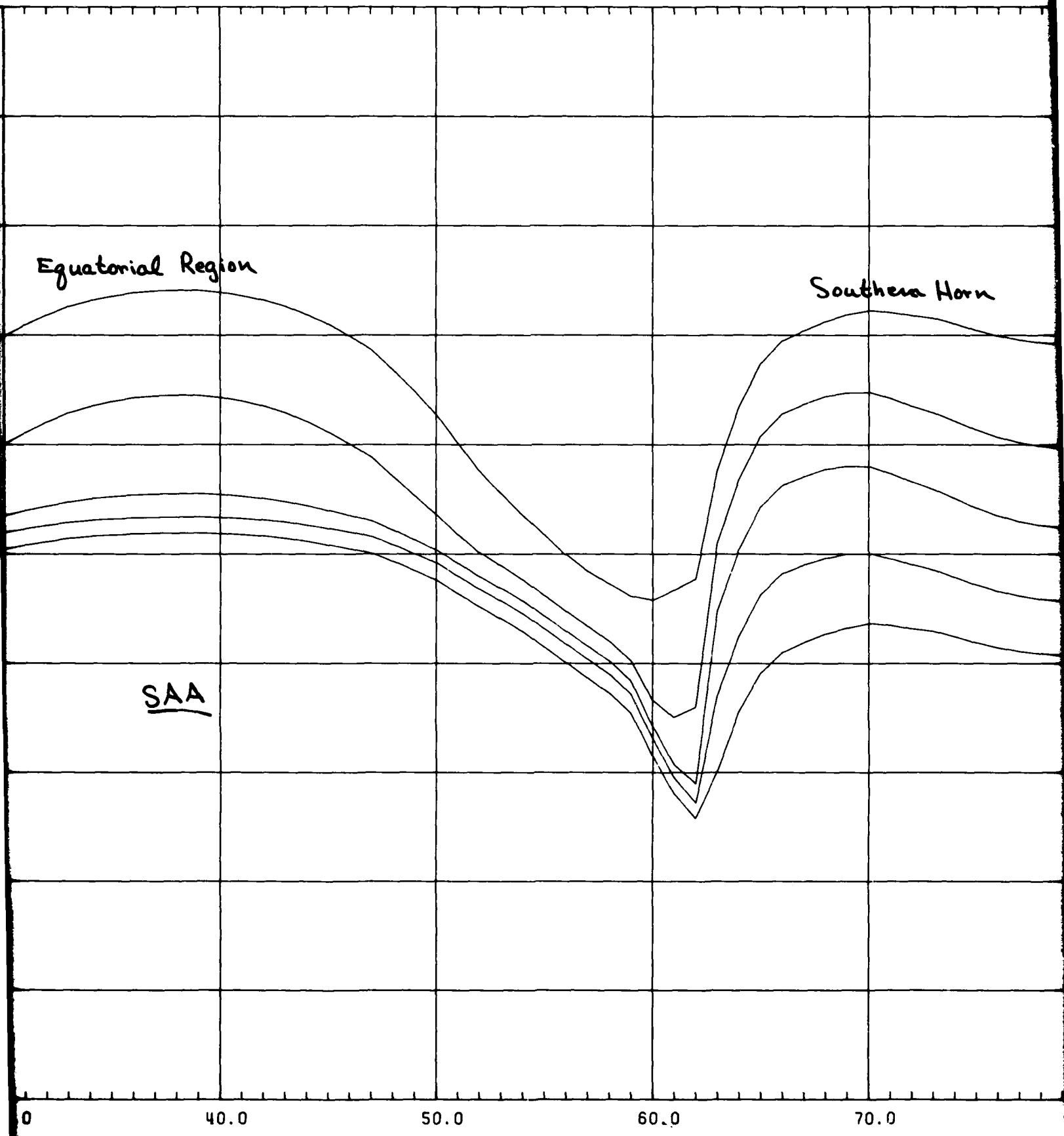
0.0 160.0 170.0 180.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



2



Equatorial Region

Southern Horn

SAA

0

40.0

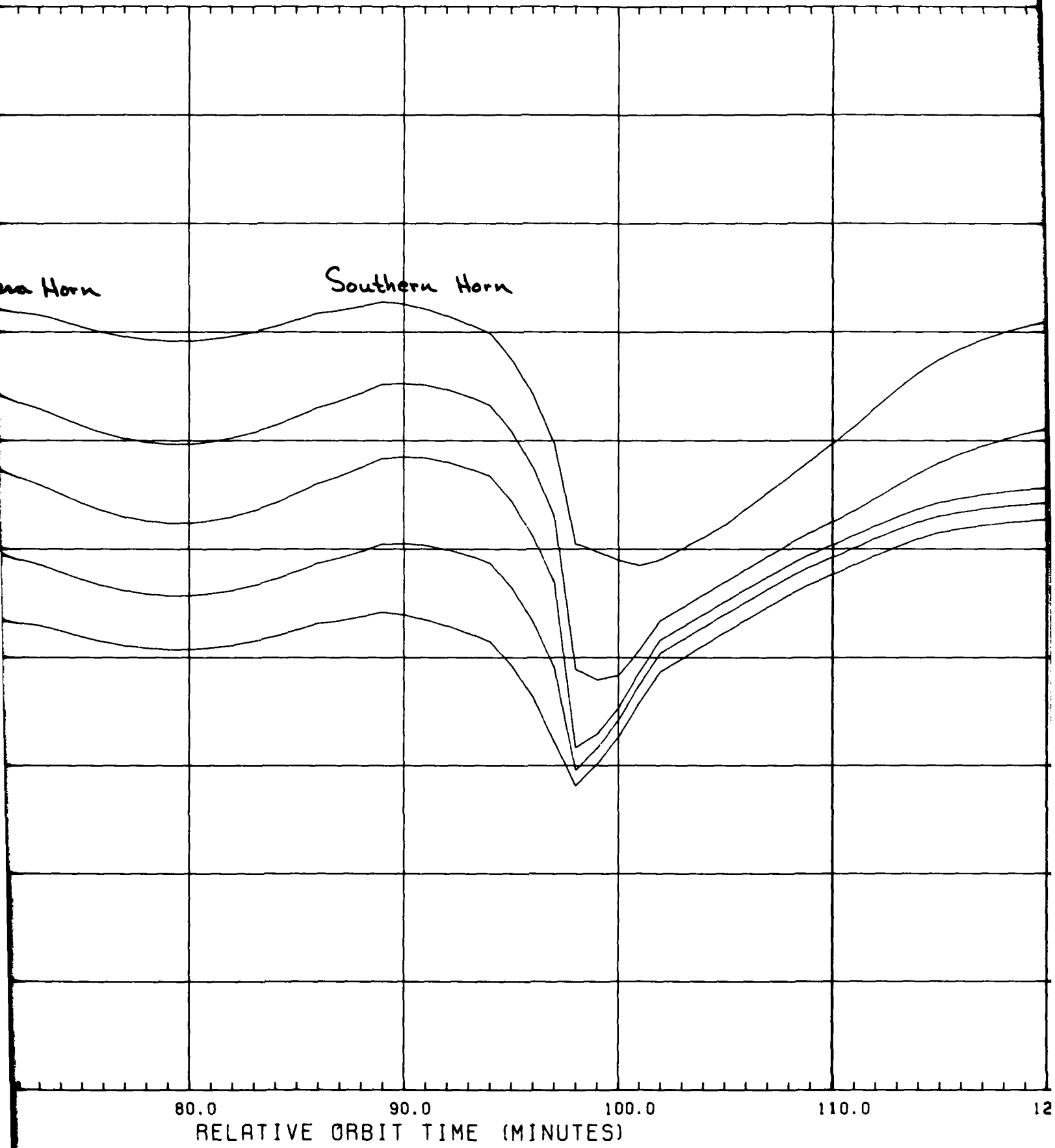
50.0

60.0

70.0

3

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



14

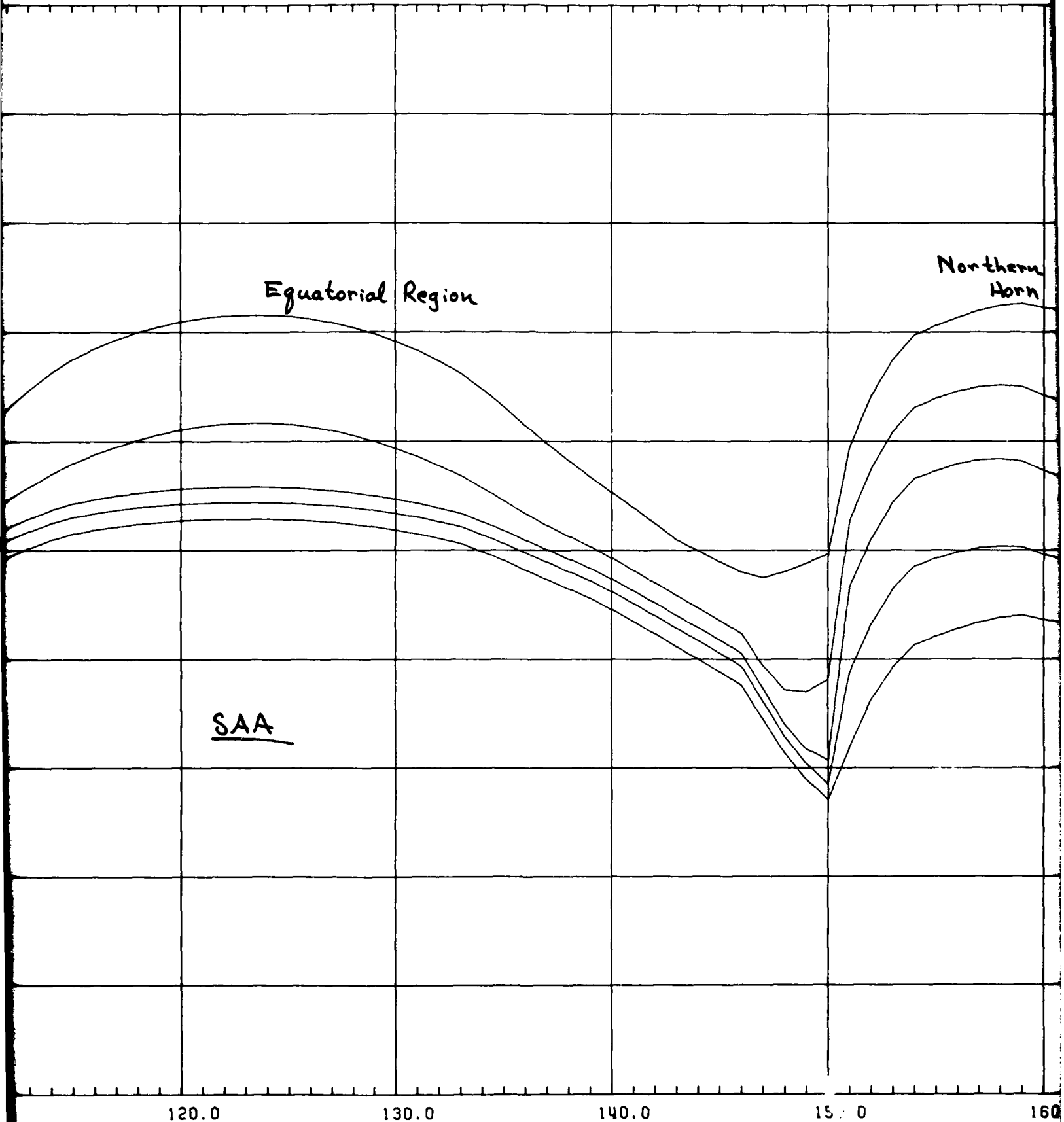




Figure 134

ORBIT: NAVELEX B  
60 DGR/3889-3889 KM

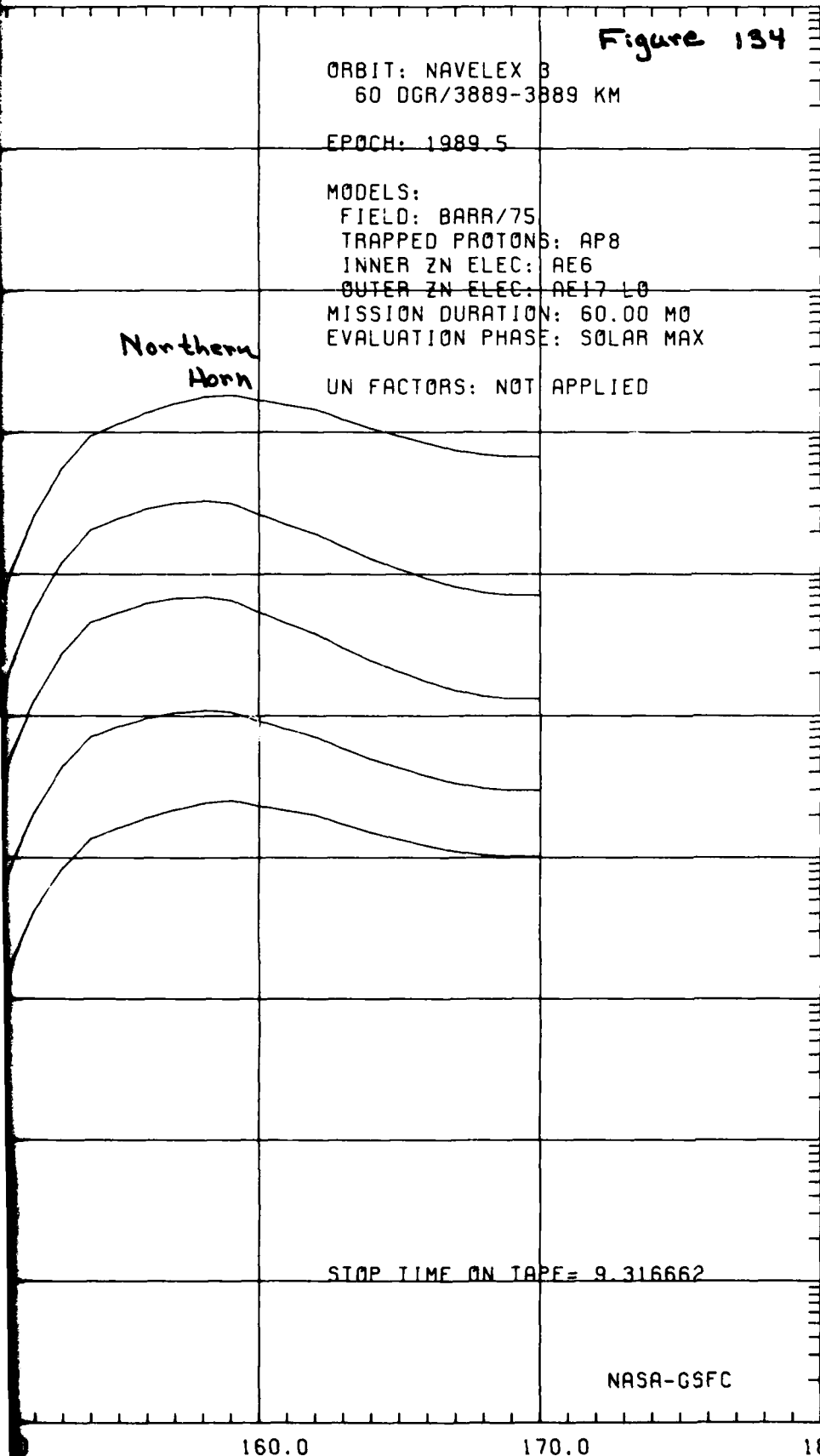
EPOCH: 1989.5

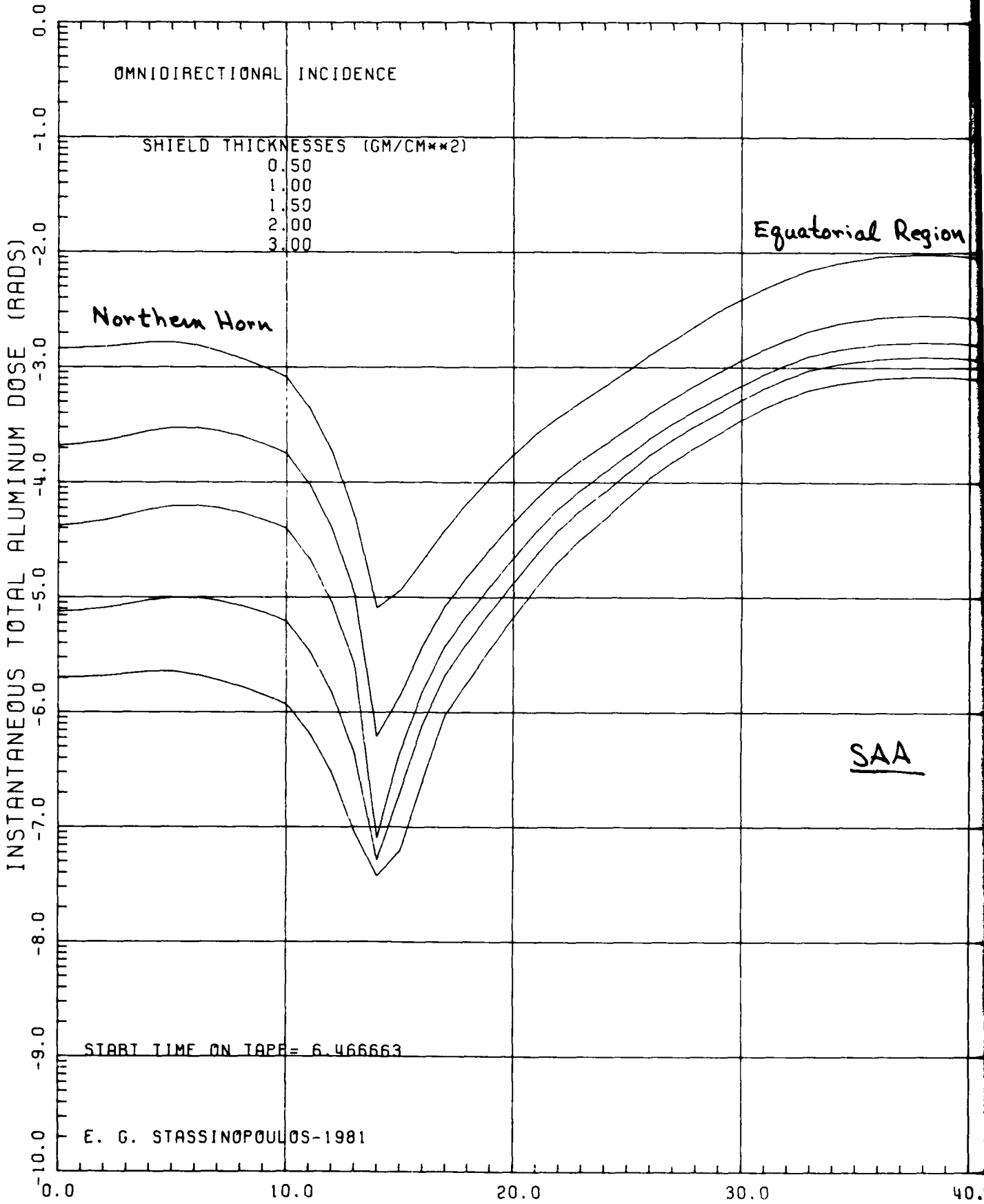
MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn





Equatorial Region

Southern Horn

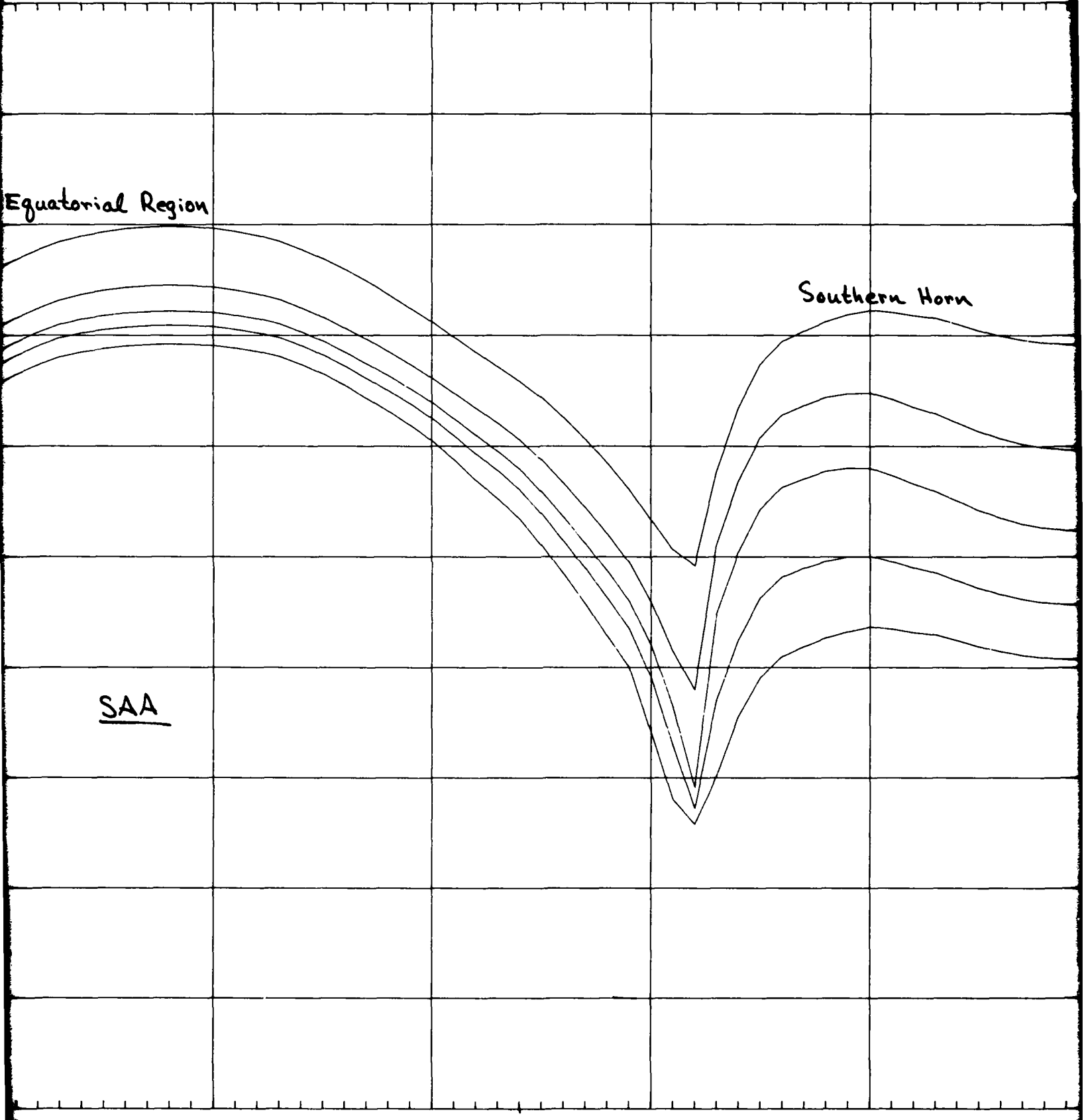
SAA

40.0

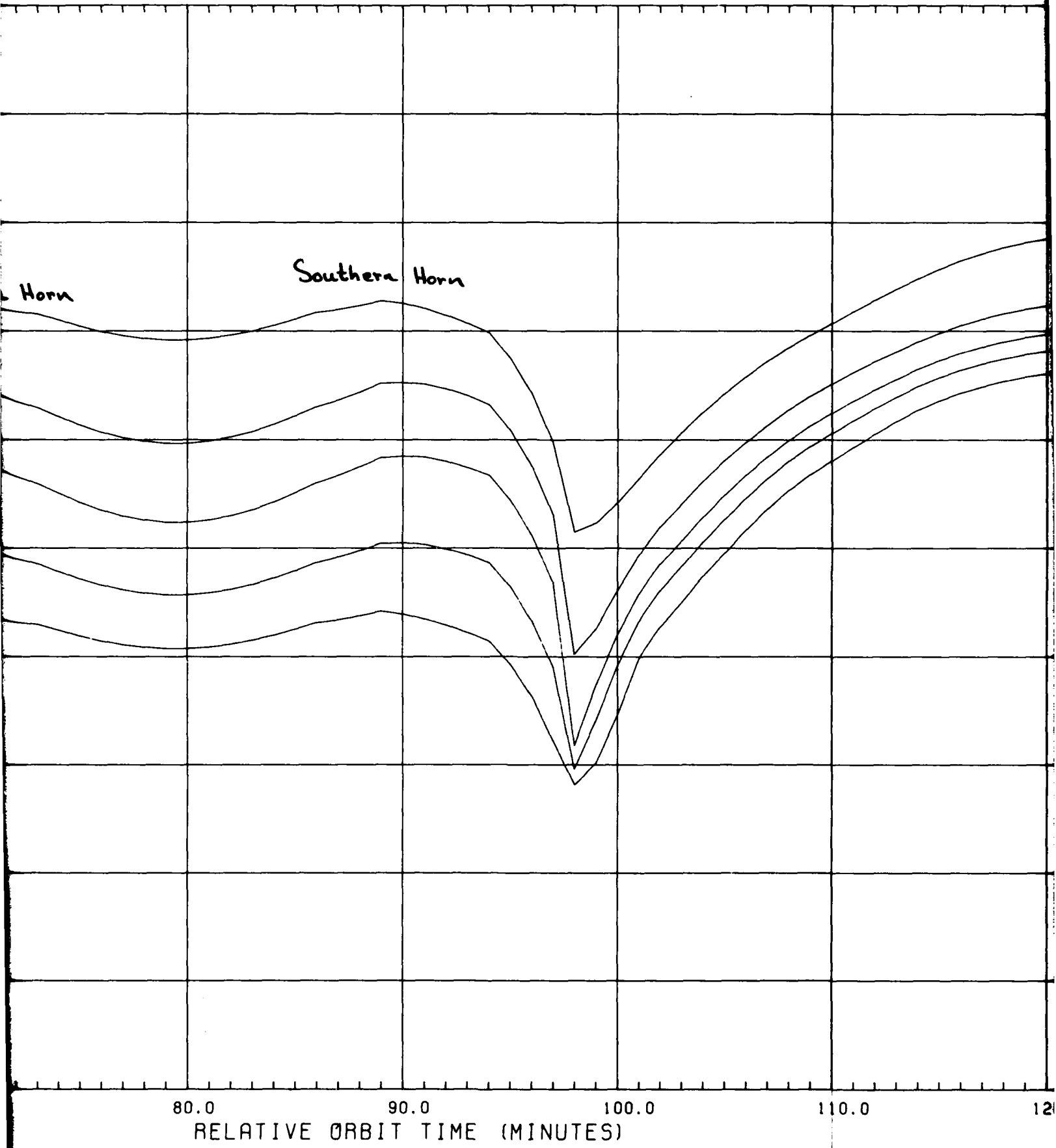
50.0

60.0

70.0



13  
DOSE IN SEMI-INFINITE ALUMINUM MEDIUM



4

Equatorial Region

Northern  
Horn

SAA

120.0

130.0

140.0

150.0

160

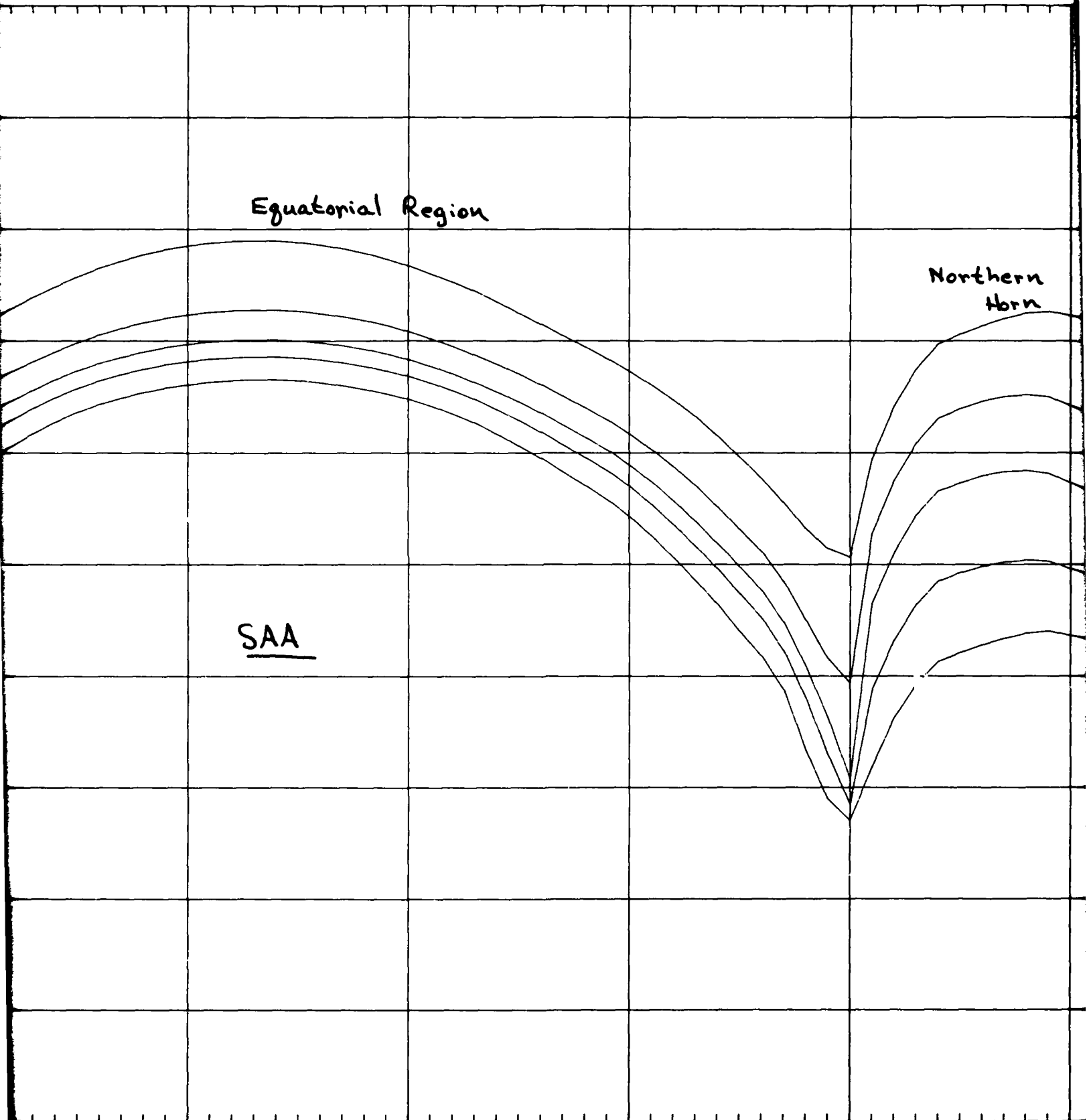


Figure 135

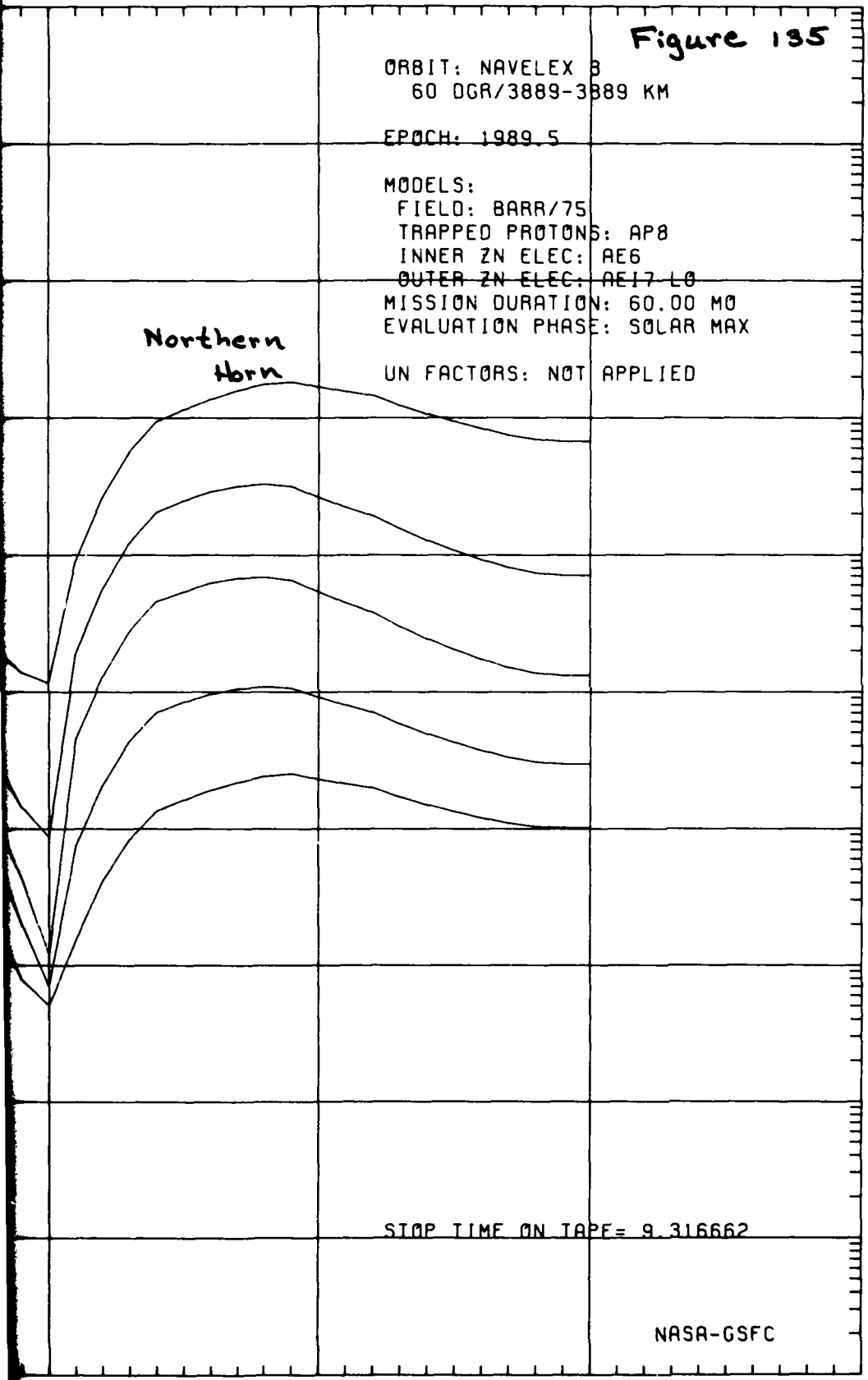
ORBIT: NAVELEX 3  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn



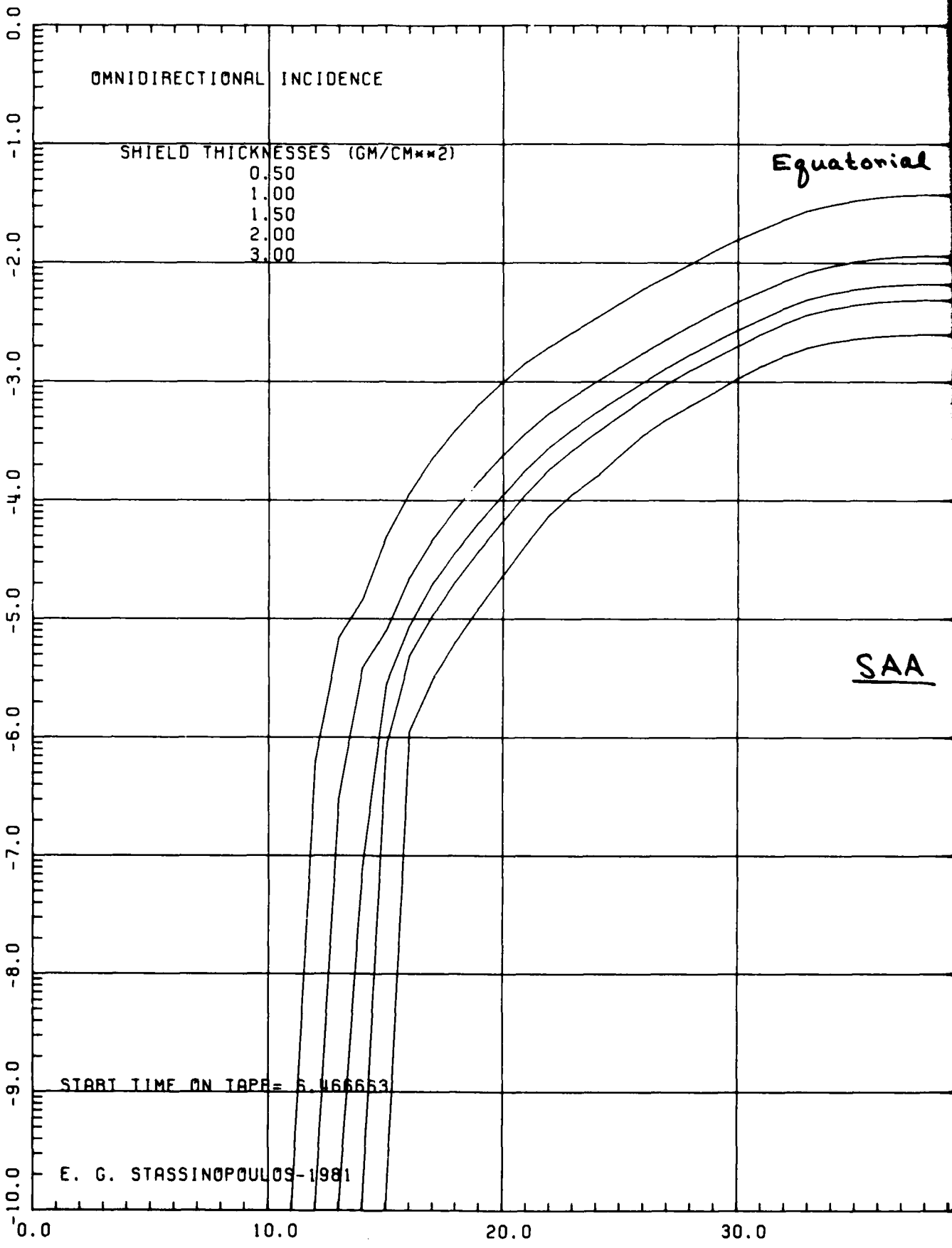
STOP TIME ON TAPE = 9.316662

NASA-GSFC

150.0                      160.0                      170.0                      180.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



201

Equatorial Region

SAA

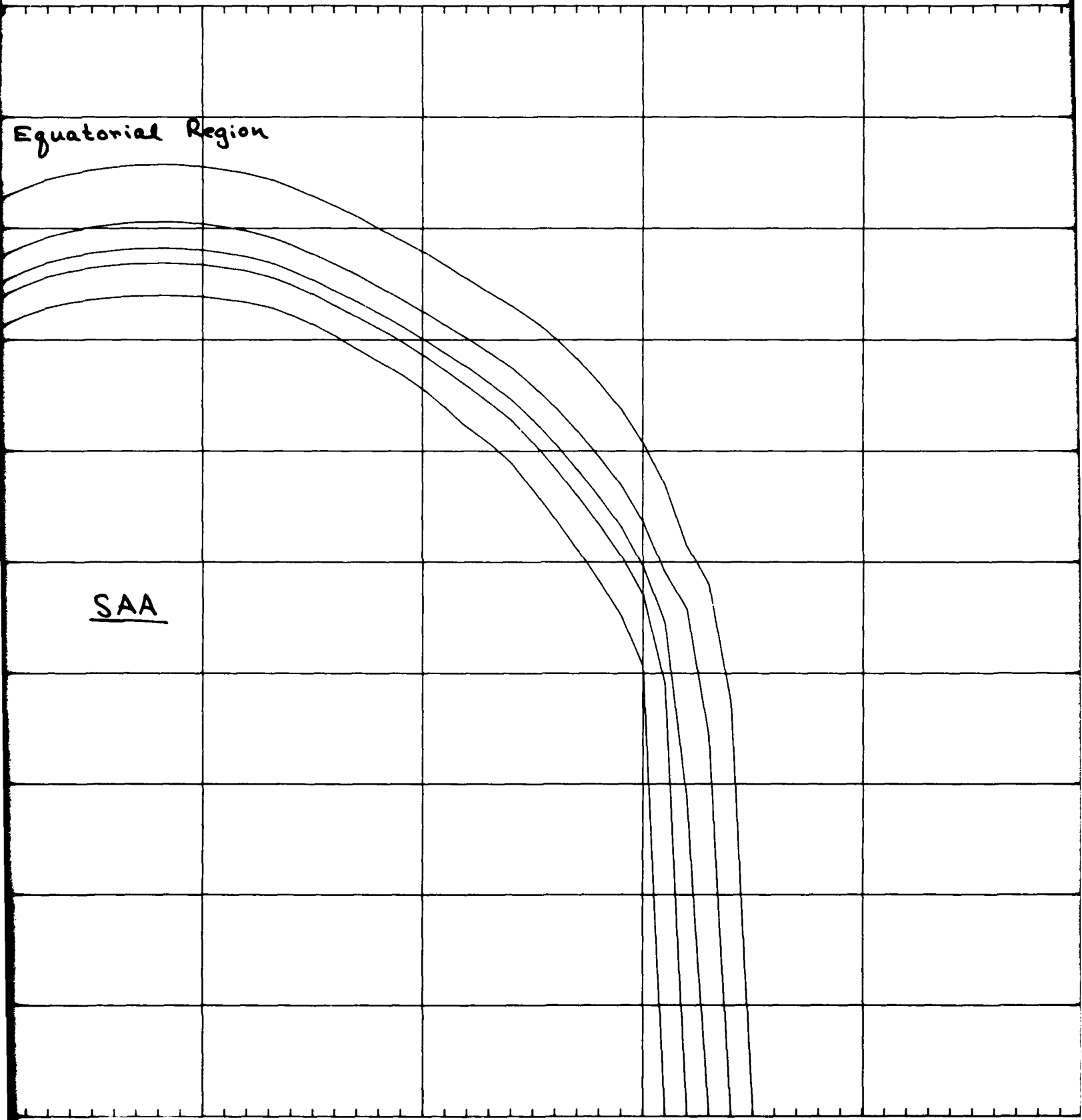
40.0

50.0

60.0

70.0

80







4'

Equatorial Region

SAA

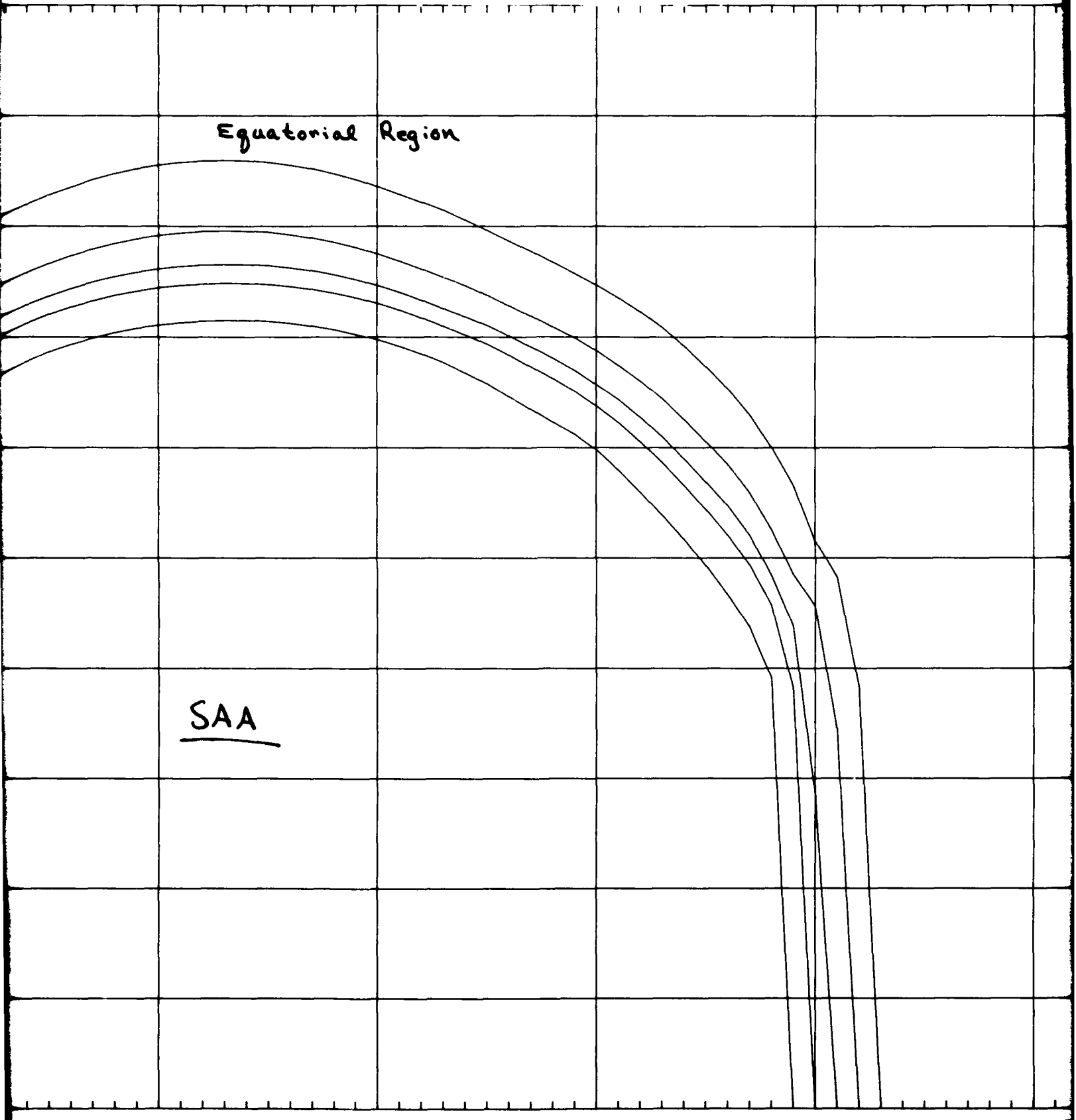
120.0

130.0

140.0

150.0

160.0



5

Figure 136

ORBIT: NAVELEX 3  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX  
  
UN FACTORS: NOT APPLIED

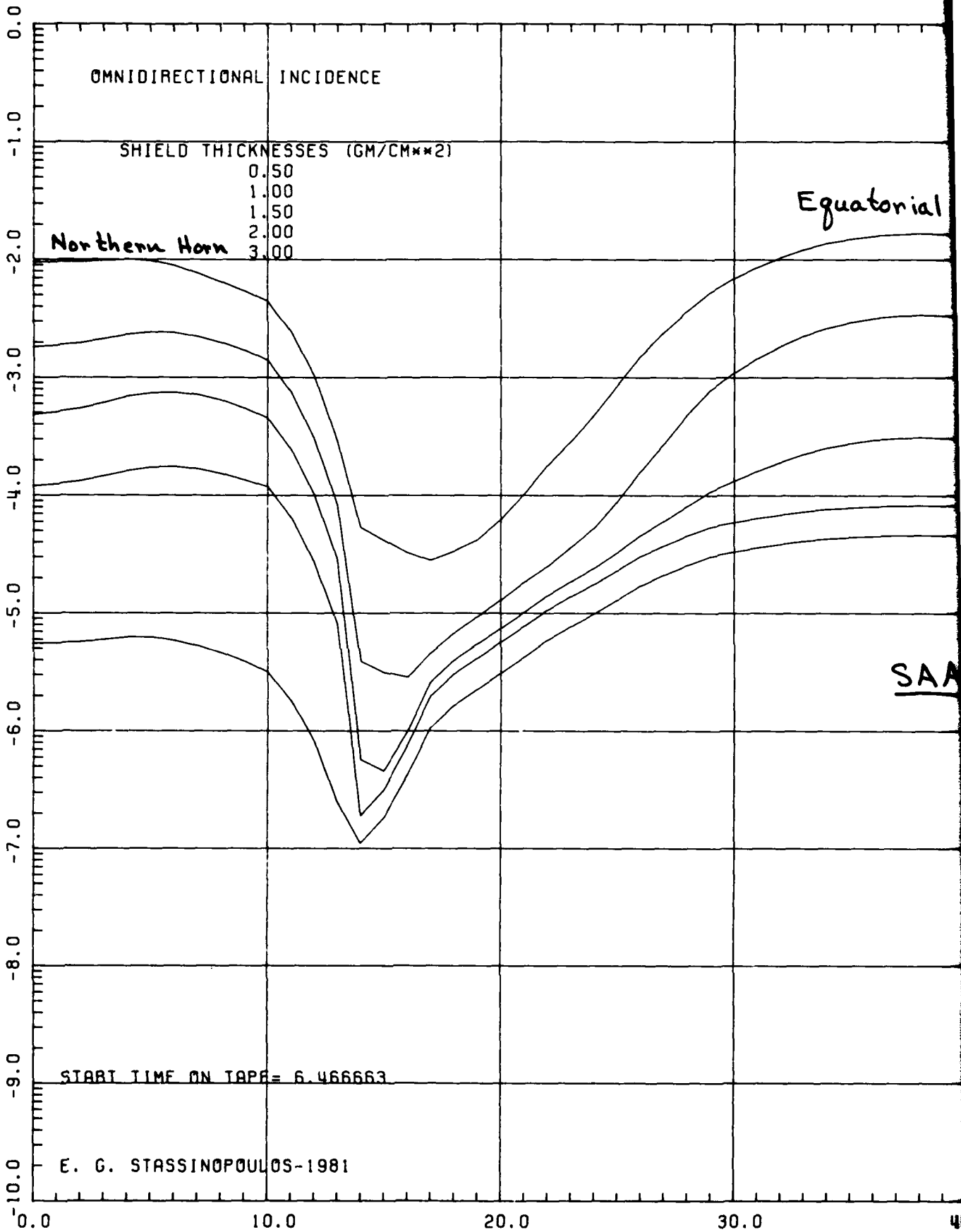
STOP TIME ON TAPE = 9.316662

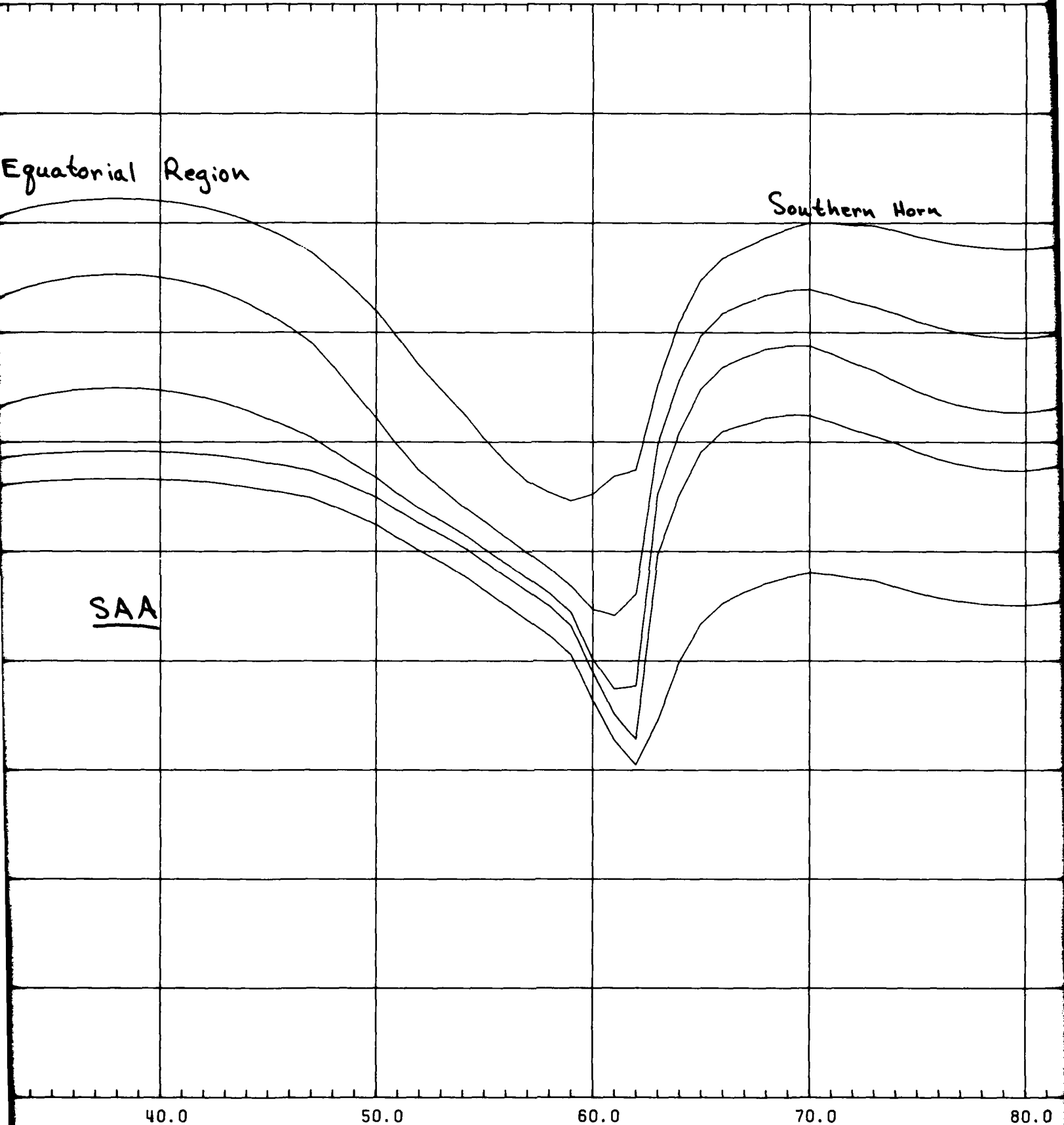
NASA-GSFC

0.0 160.0 170 0 180.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)





Equatorial Region

Southern Horn

SAA

40.0

50.0

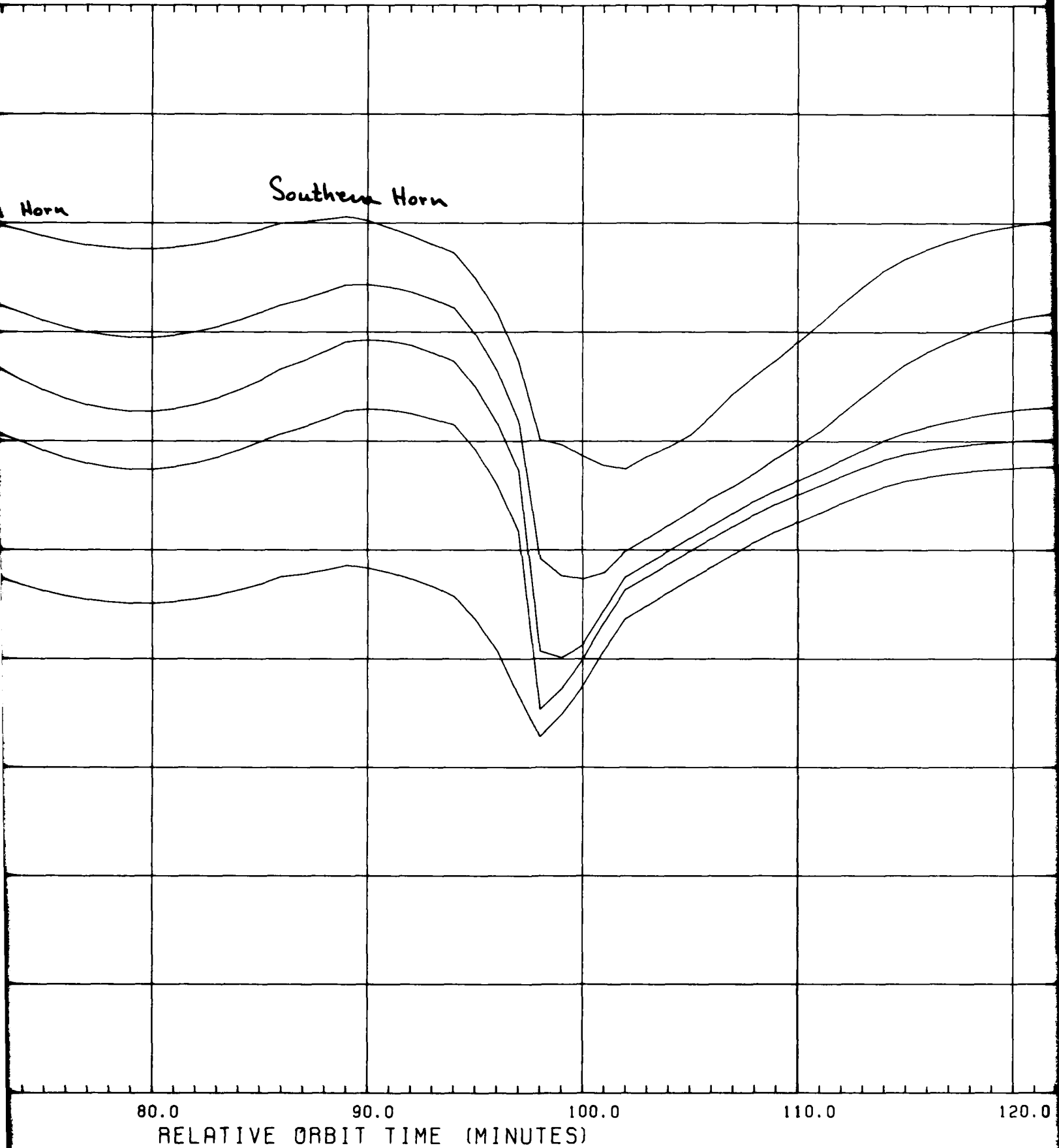
60.0

70.0

80.0

REL

3  
DOSE AT CENTER OF ALUMINUM SPHERES



4

Equatorial Region

Northern  
Horn

SAA

120.0

130.0

140.0

150.0

160.0

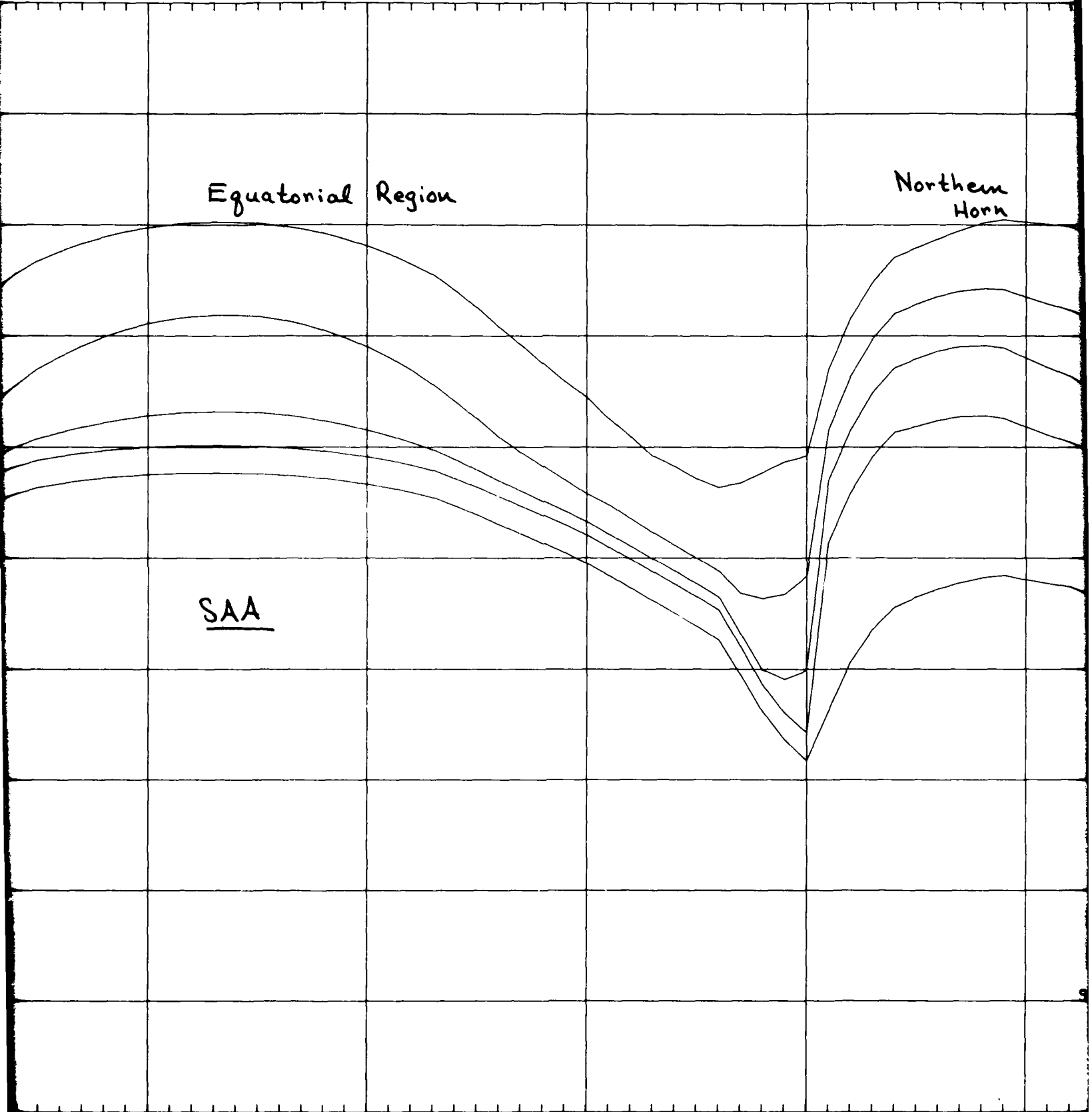


Figure 137

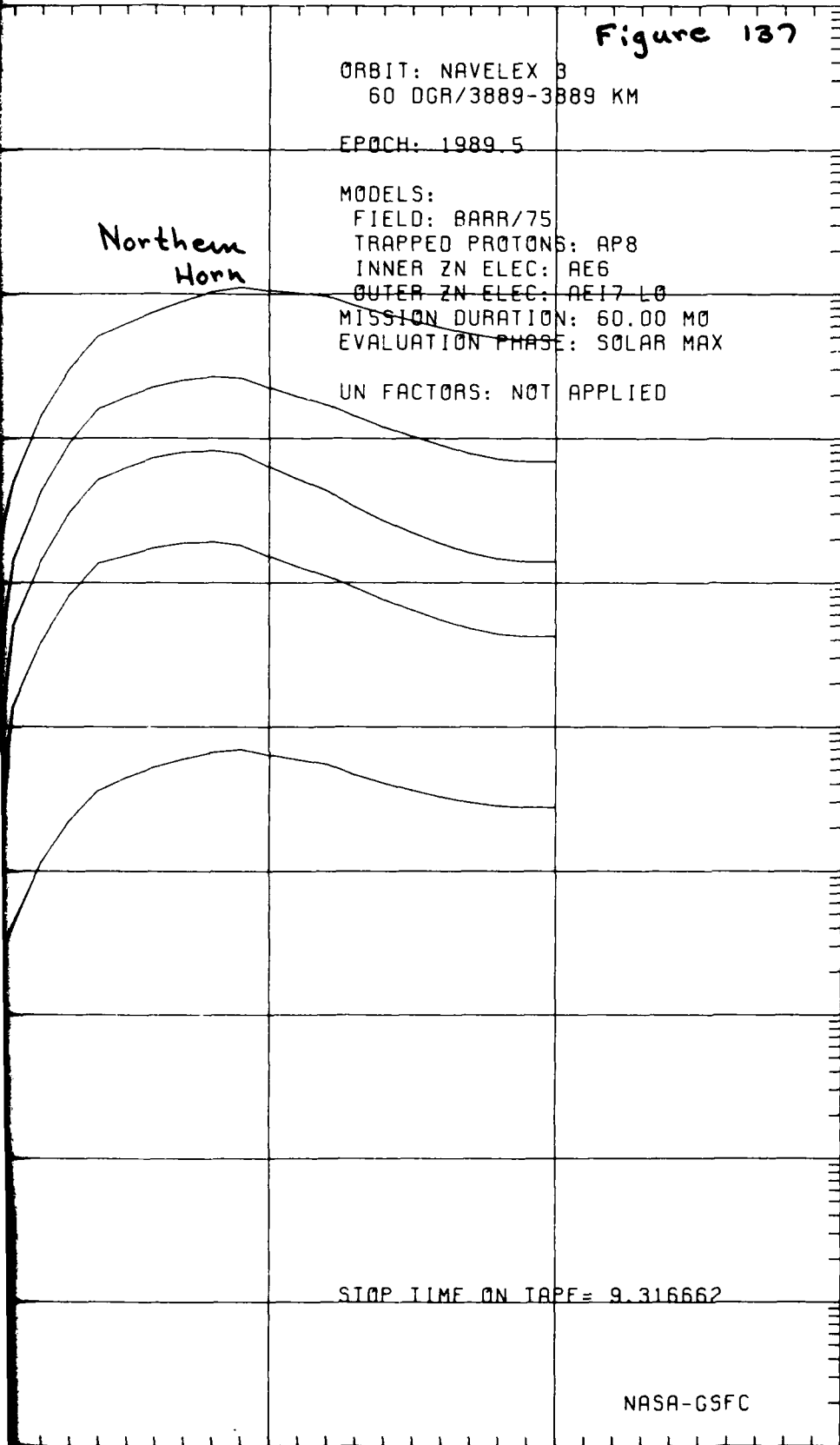
ORBIT: NAVELEX 3  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn

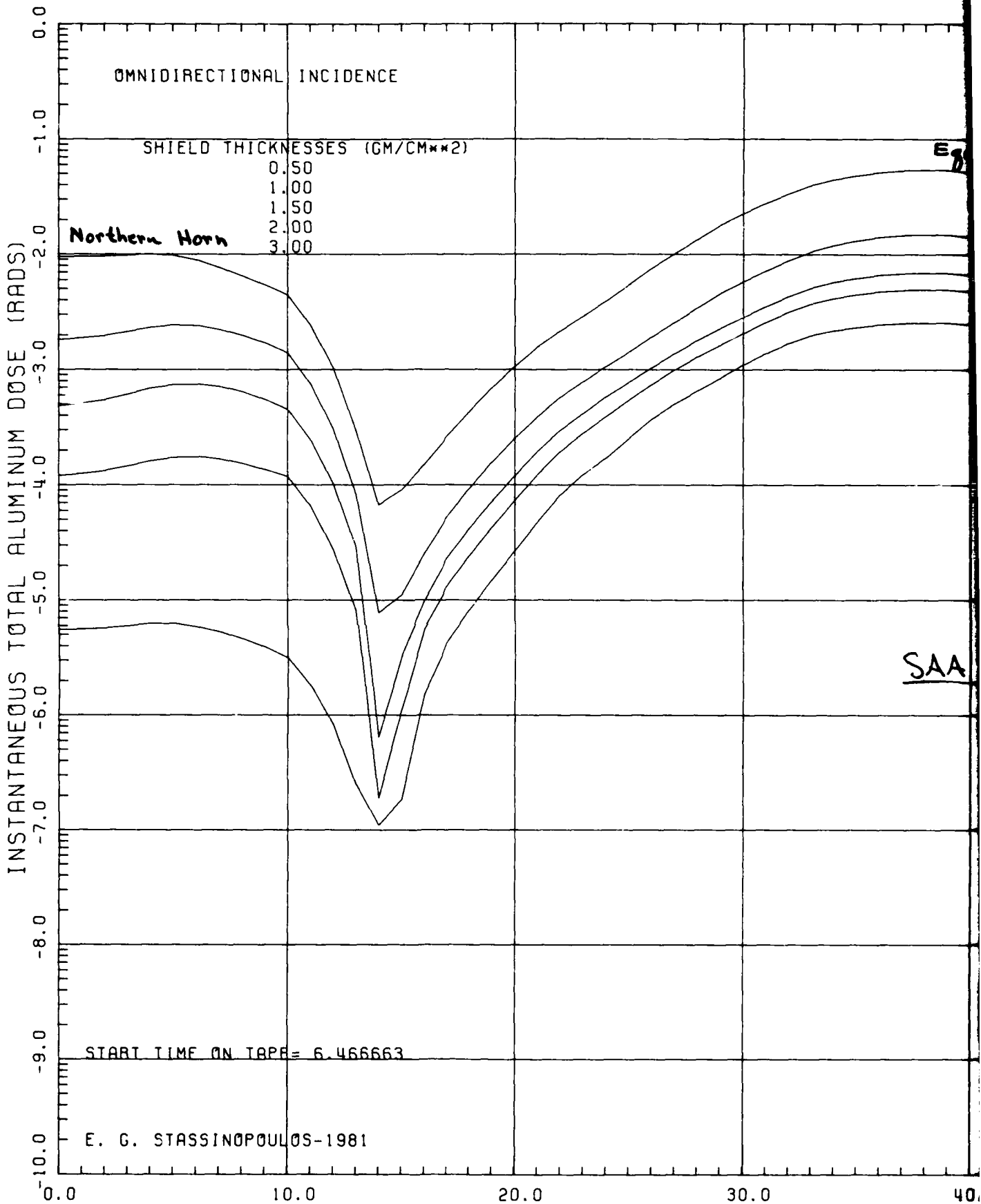


STOP TIME ON TAPE = 9.316662

NASA-GSFC

0 160.0 170.0 180.0





2

Equatorial Region

Southern Hem

SAA

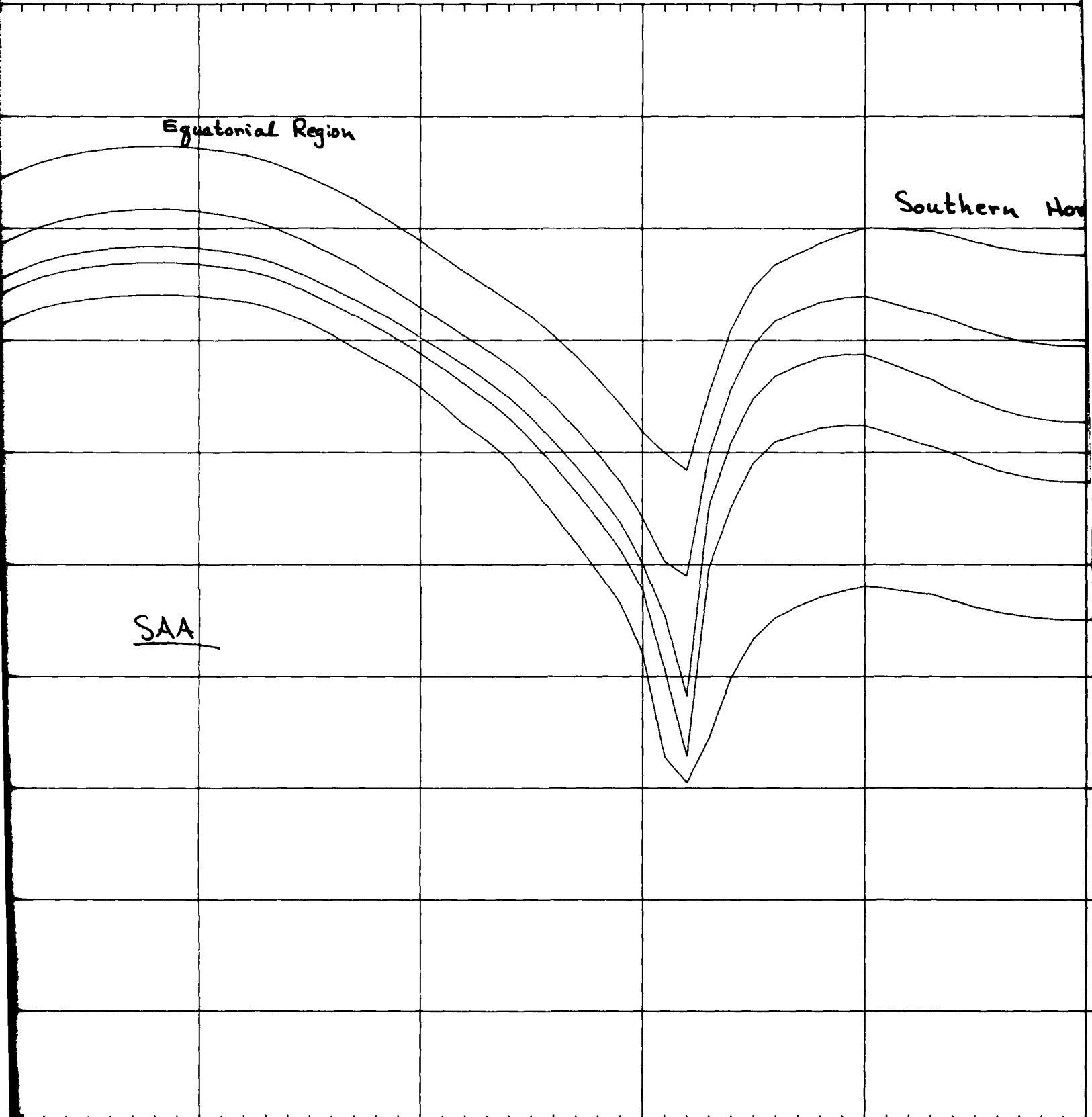
40.0

50.0

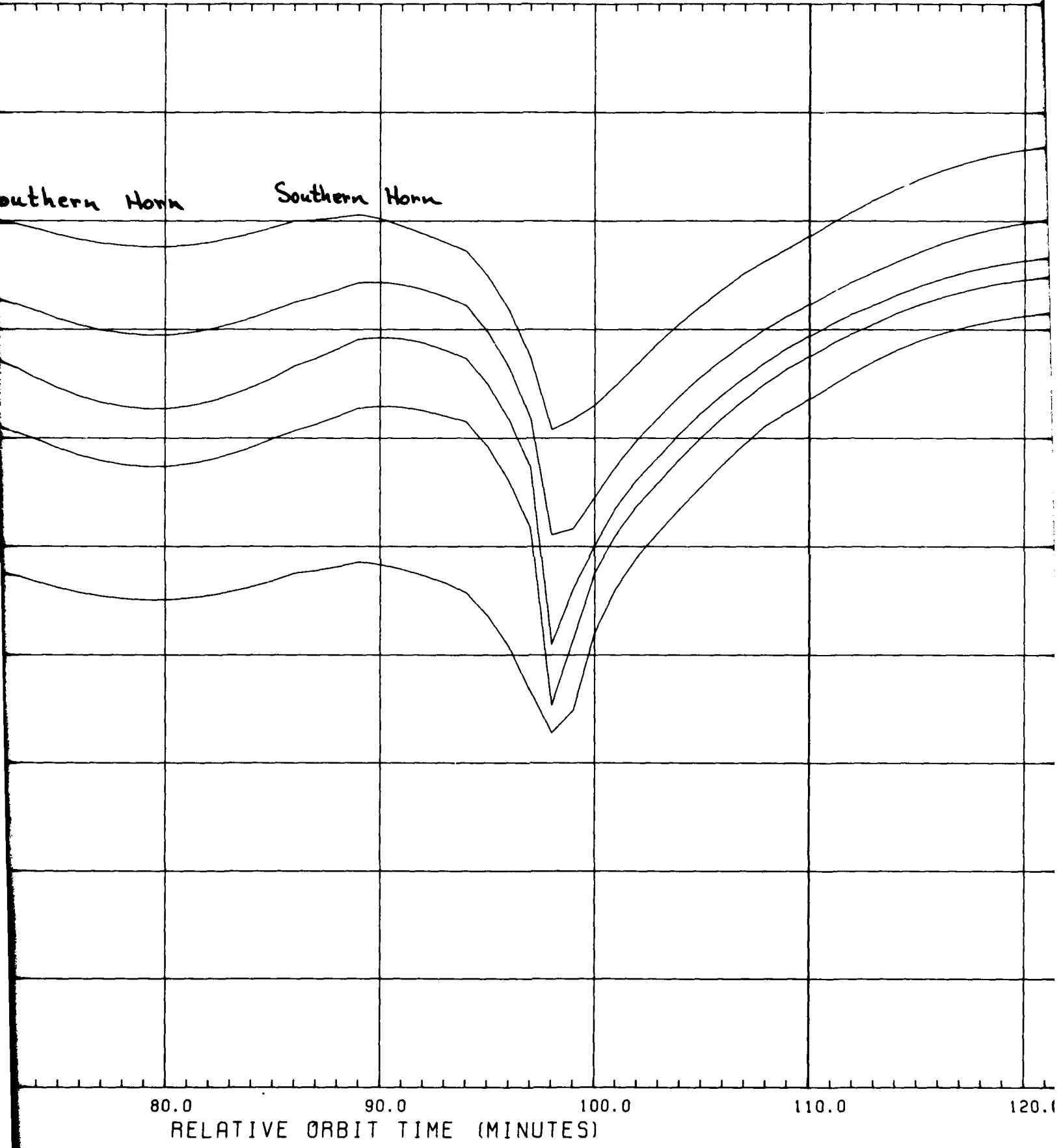
60.0

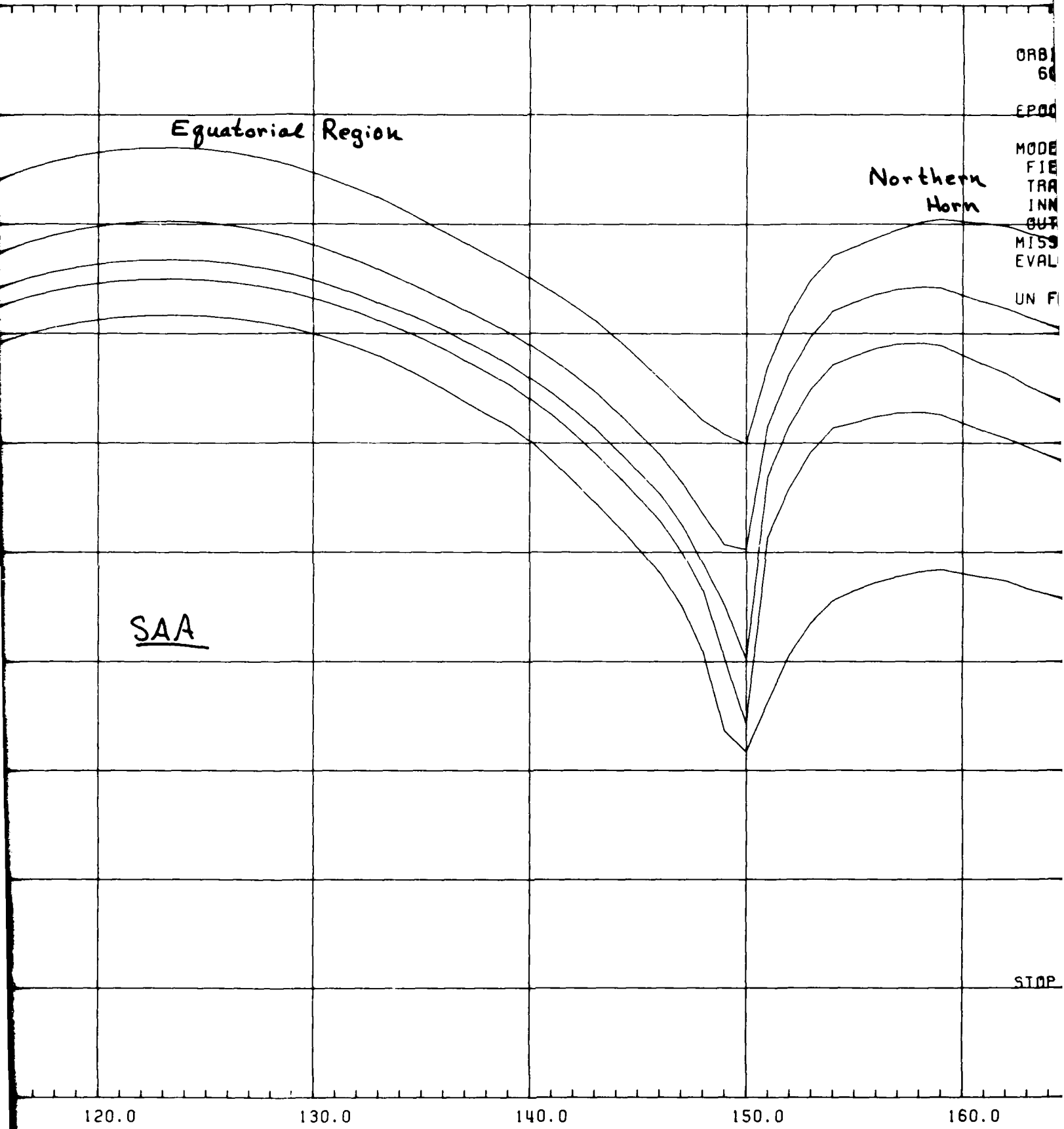
70.0

80



12  
DOSE AT CENTER OF ALUMINUM SPHERES





Equatorial Region

Northern  
Horn

SAA

ORB  
60  
EP00  
MODE  
FIE  
TRA  
INN  
GUR  
MISS  
EVAL  
UN FI

STOP

120.0

130.0

140.0

150.0

160.0

Figure 138

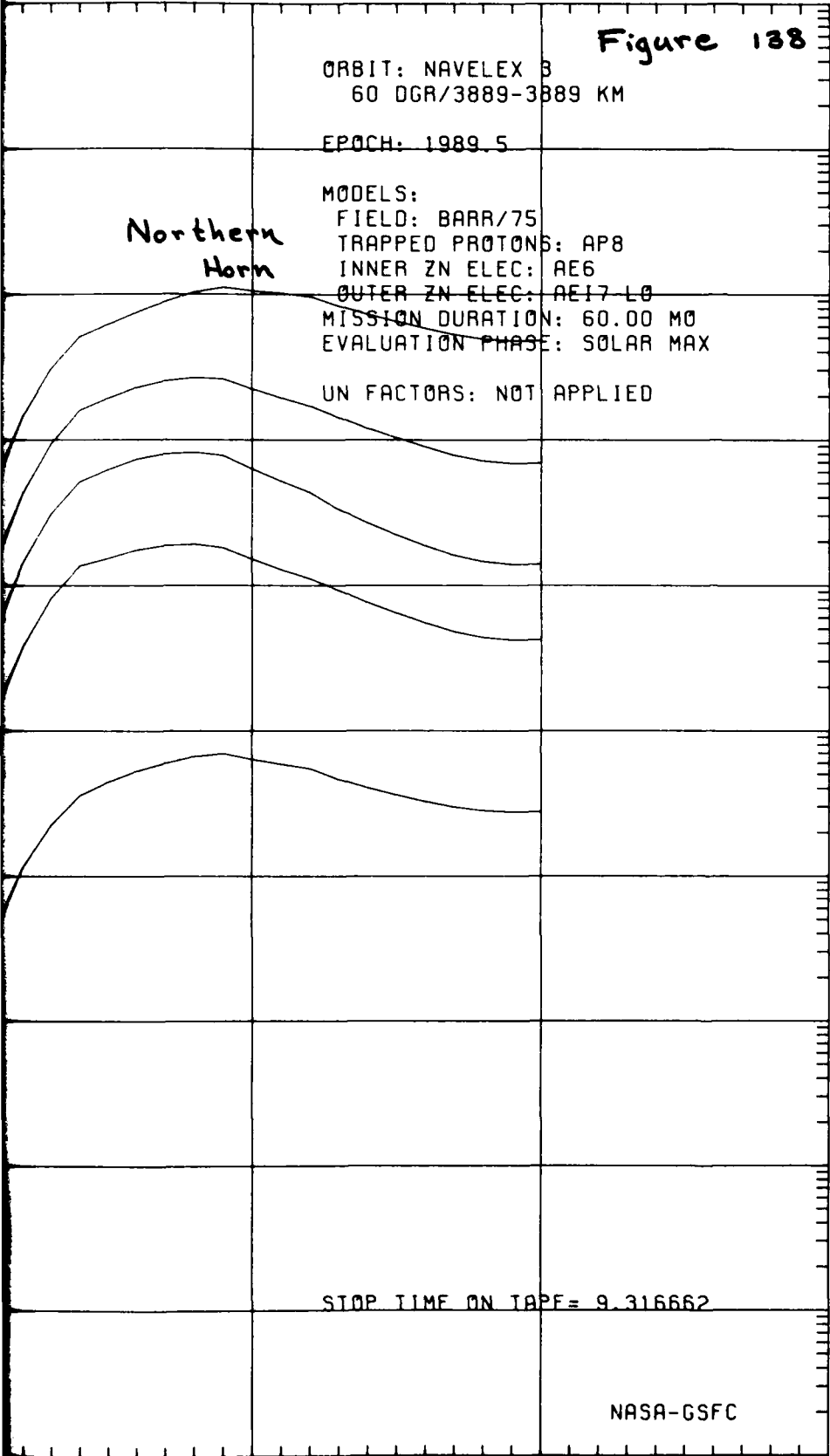
ORBIT: NAVELEX B  
60 DGR/3889-3889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn



STOP TIME ON TAPE = 9.316662

NASA-GSFC

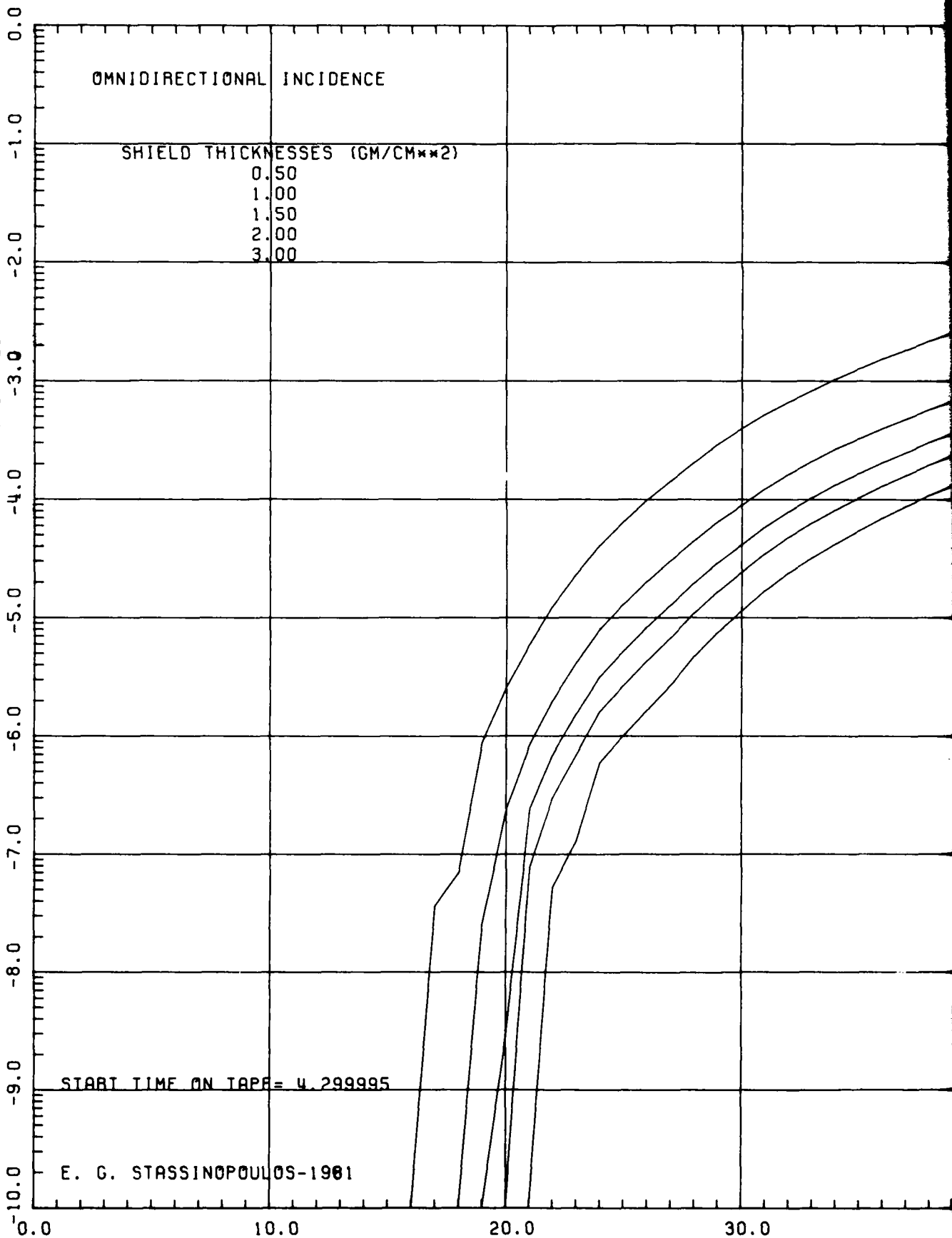
160.0

170.0

180.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



10

Equatorial Region

SAA

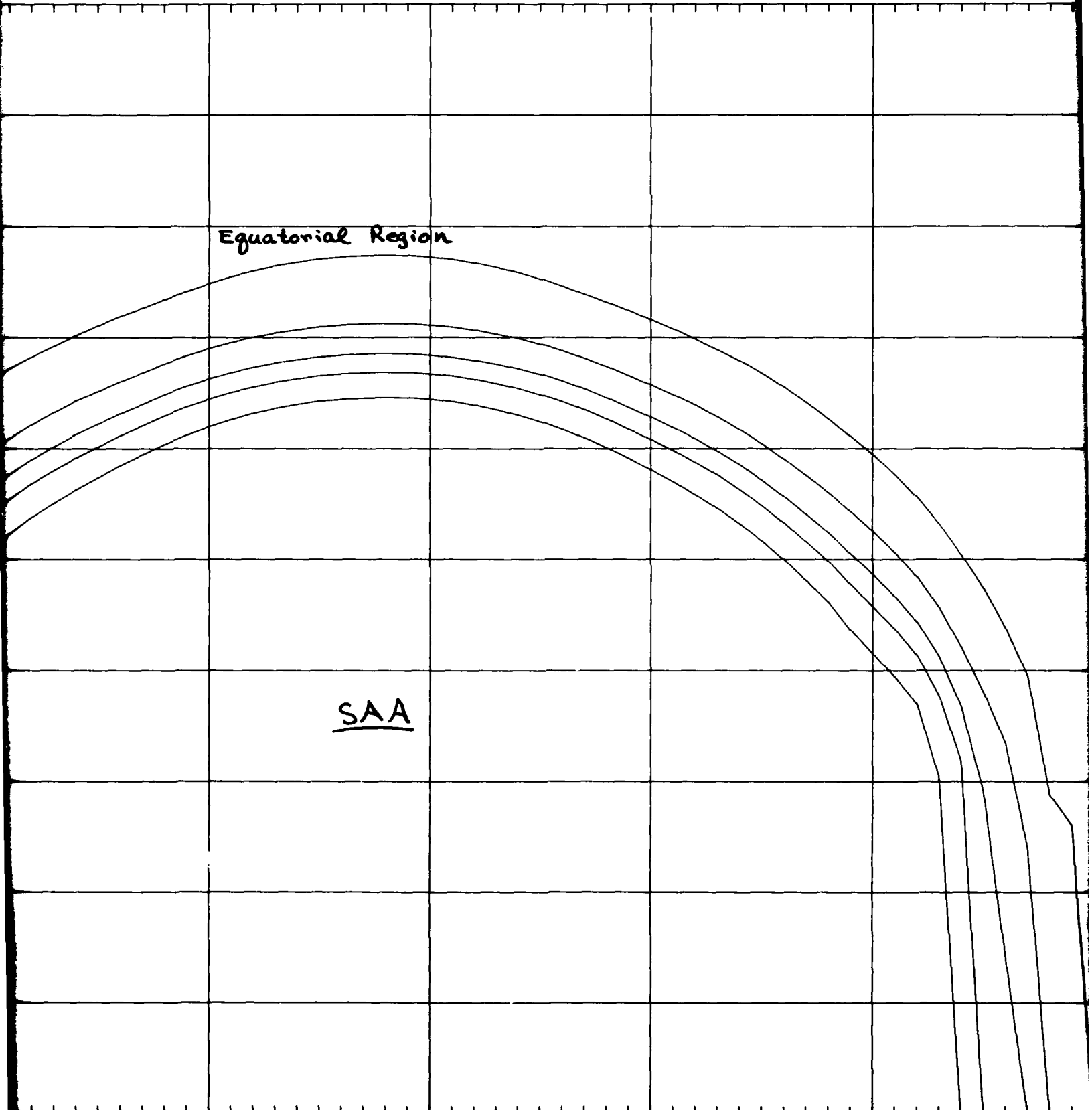
40.0

50.0

60.0

70.0

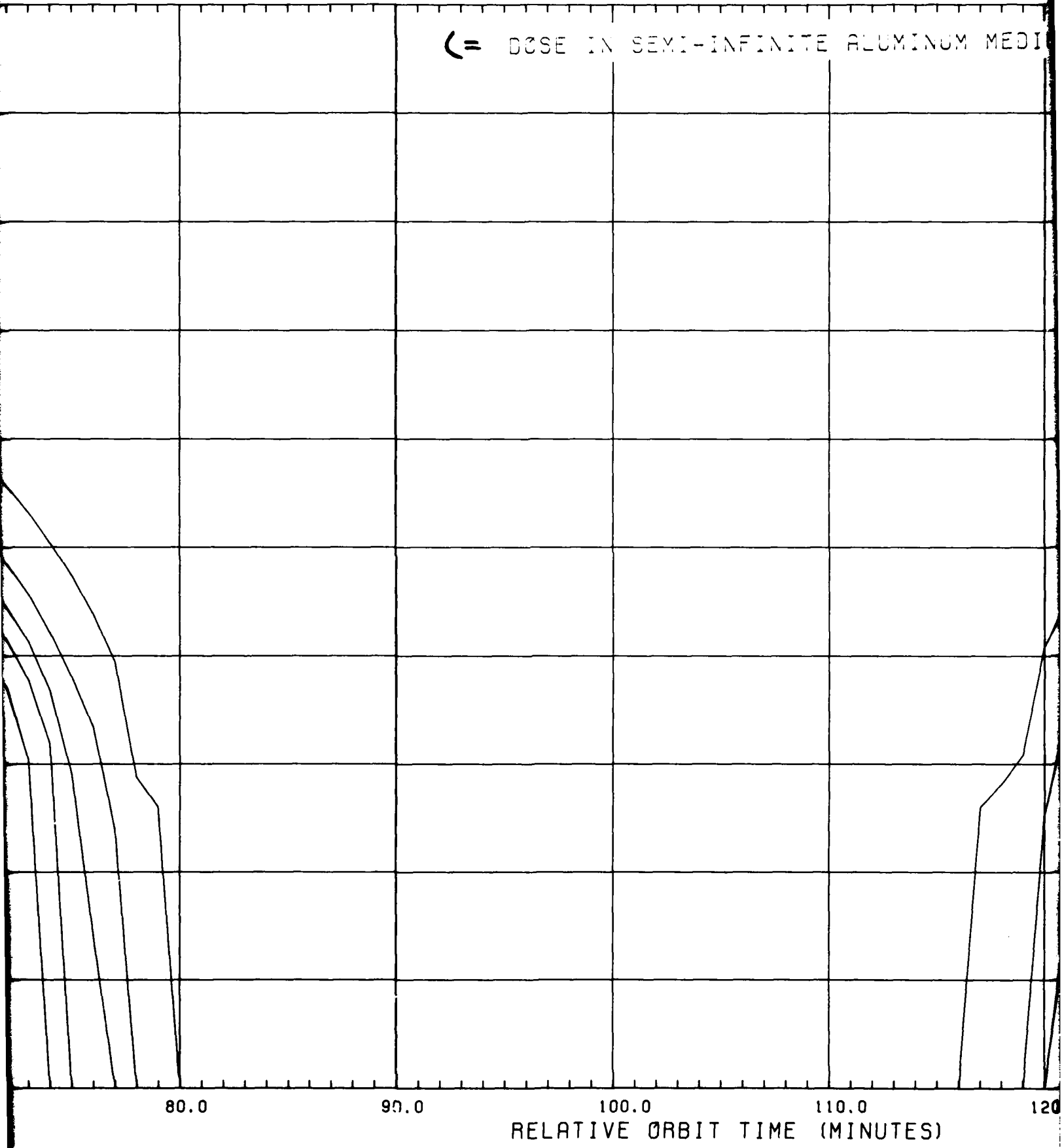
8



3

DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM

$\leftarrow$  DOSE IN SEMI-INFINITE ALUMINUM MEDIUM





4

# STEEL ALUMINUM SLAB SHIELDS

ALUMINUM MEDIUM)

Equatorial Re

SAA

120.0

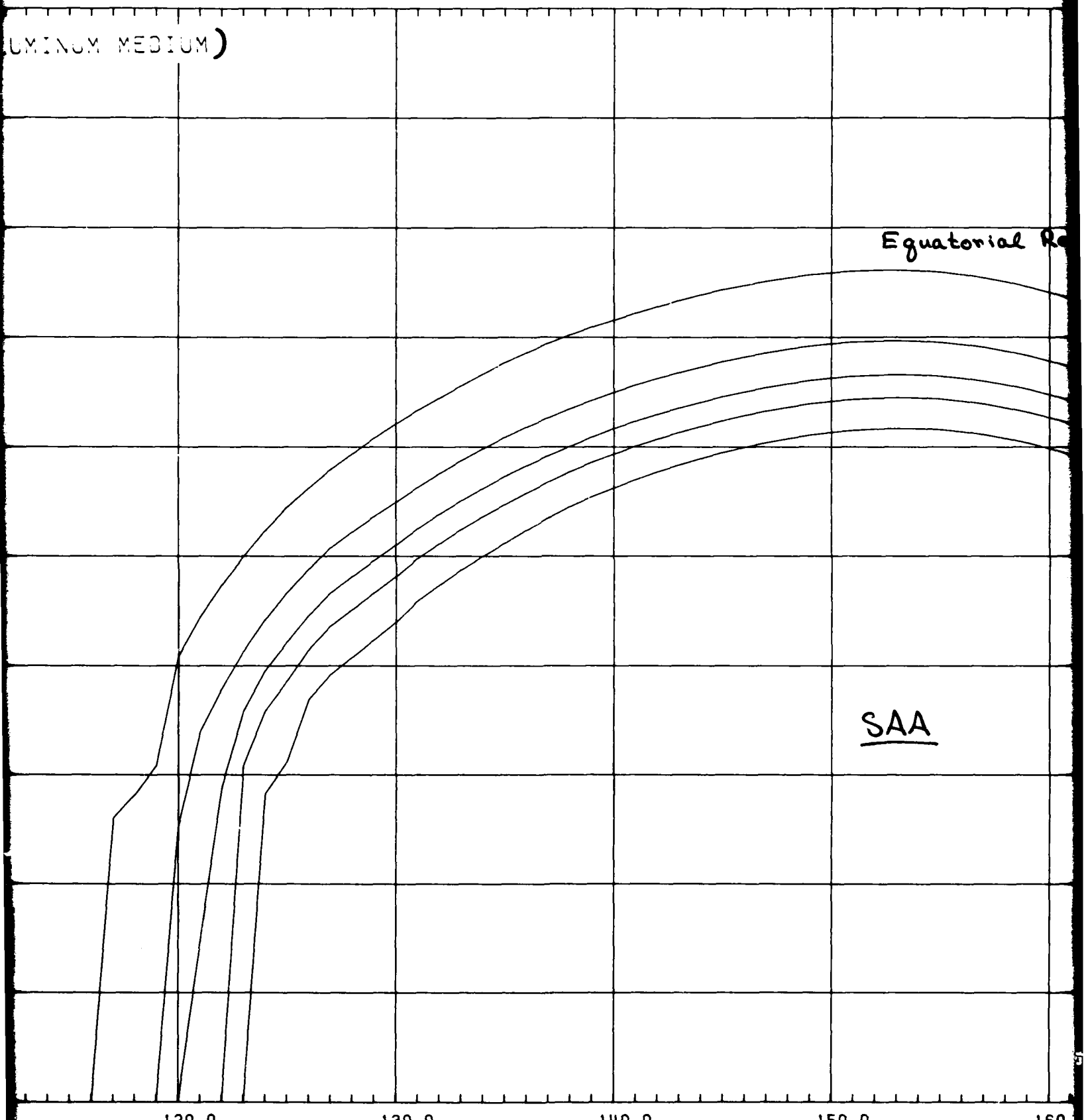
130.0

140.0

150.0

160.0

TES)



5

ORBIT: NAVELE  
60 DGR/5186

EPOCH: 1989.5

MODELS:  
FIELD: BARR/  
TRAPPED PROT  
INNER ZN ELE  
OUTER ZN ELE  
MISSION DURAT  
EVALUATION PH

UN FACTORS: N

Equatorial Region

SAA

STOP TIME ON T

0 160.0 170.0 180.0 190.0

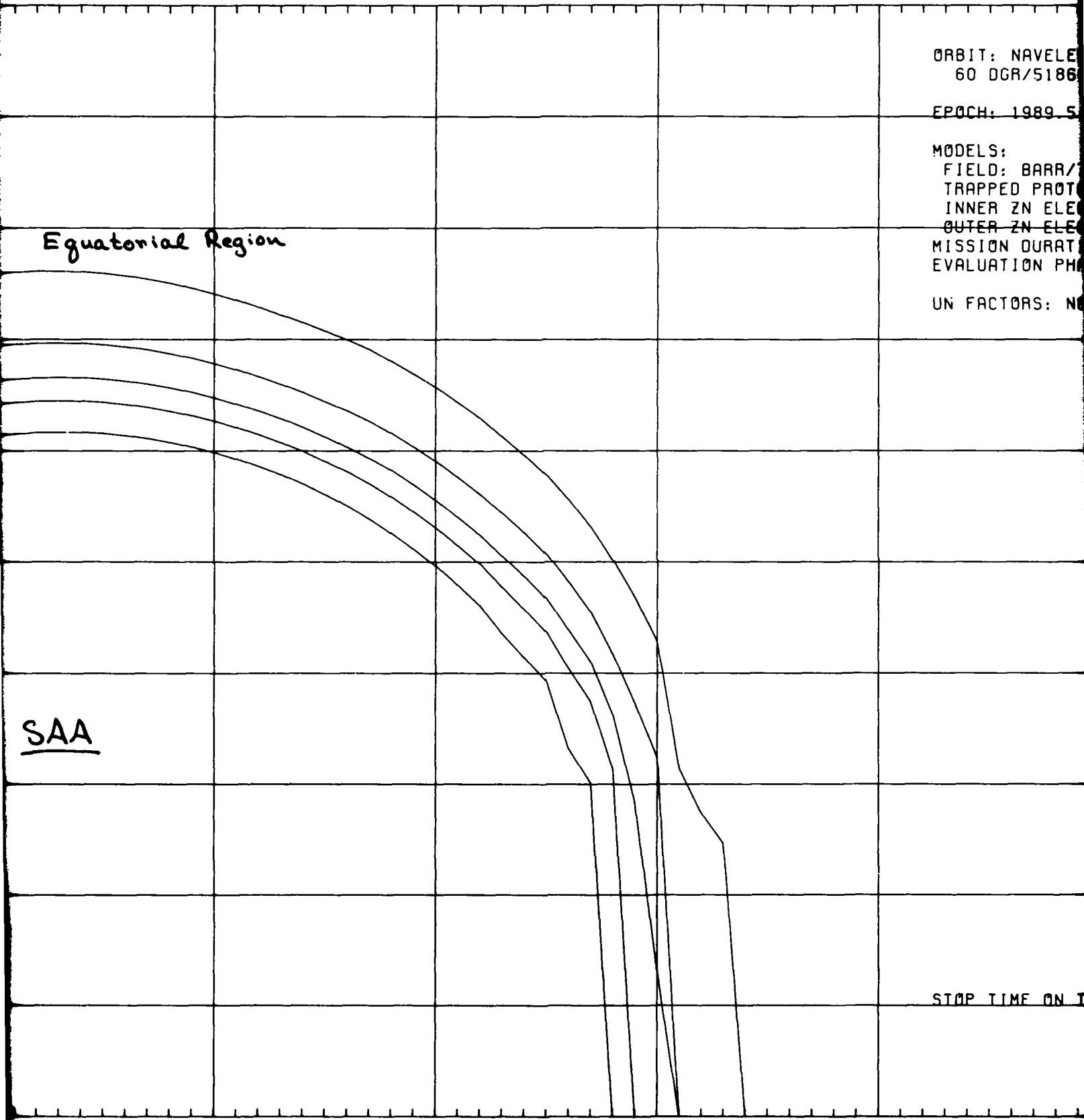


Figure 139

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: A66

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 M0

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 7.716662

NASA-GSFC

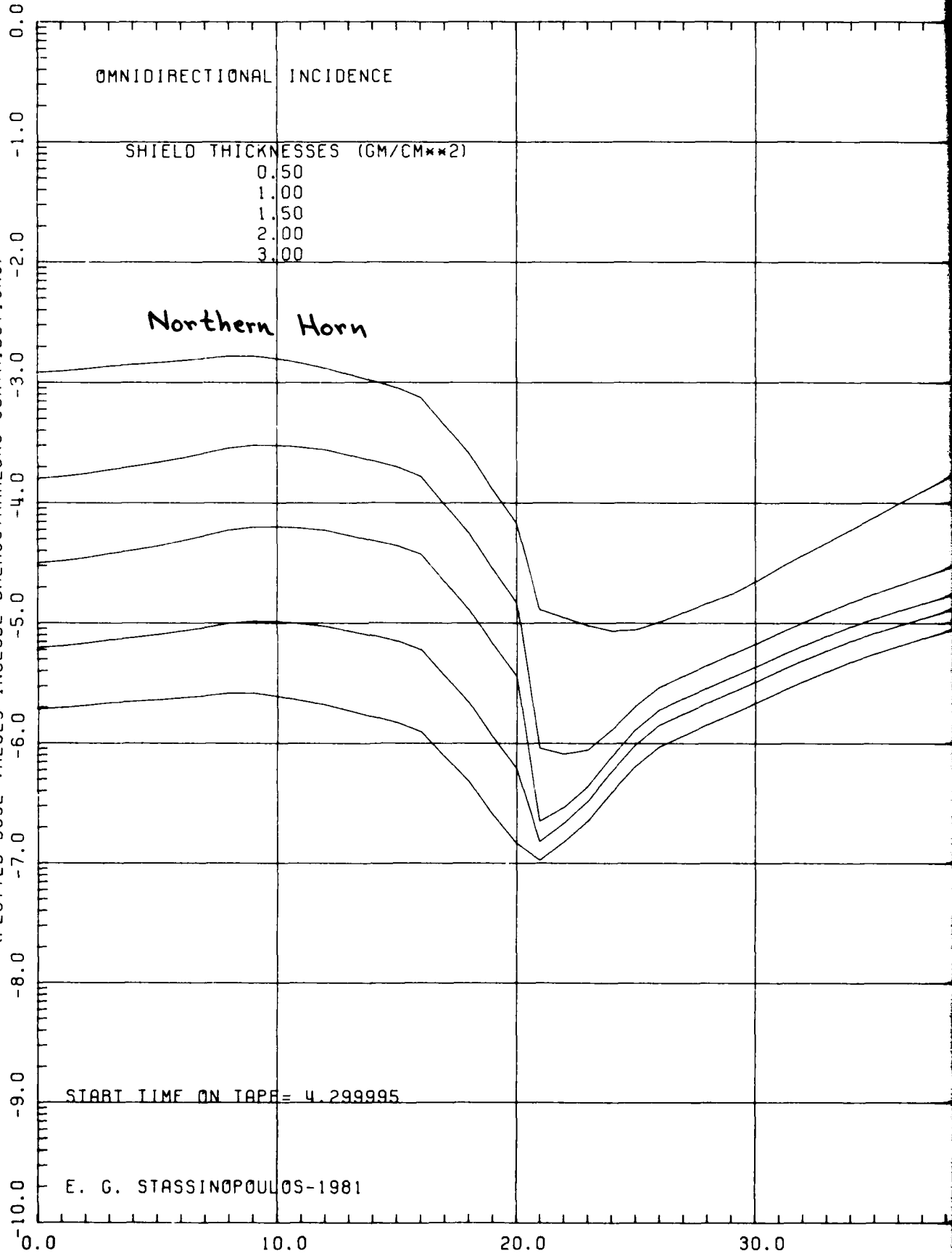
90.0

200.0

210.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



OMNIDIRECTIONAL INCIDENCE

SHIELD THICKNESSES (GM/CM\*\*2)  
0.50  
1.00  
1.50  
2.00  
3.00

Northern Horn

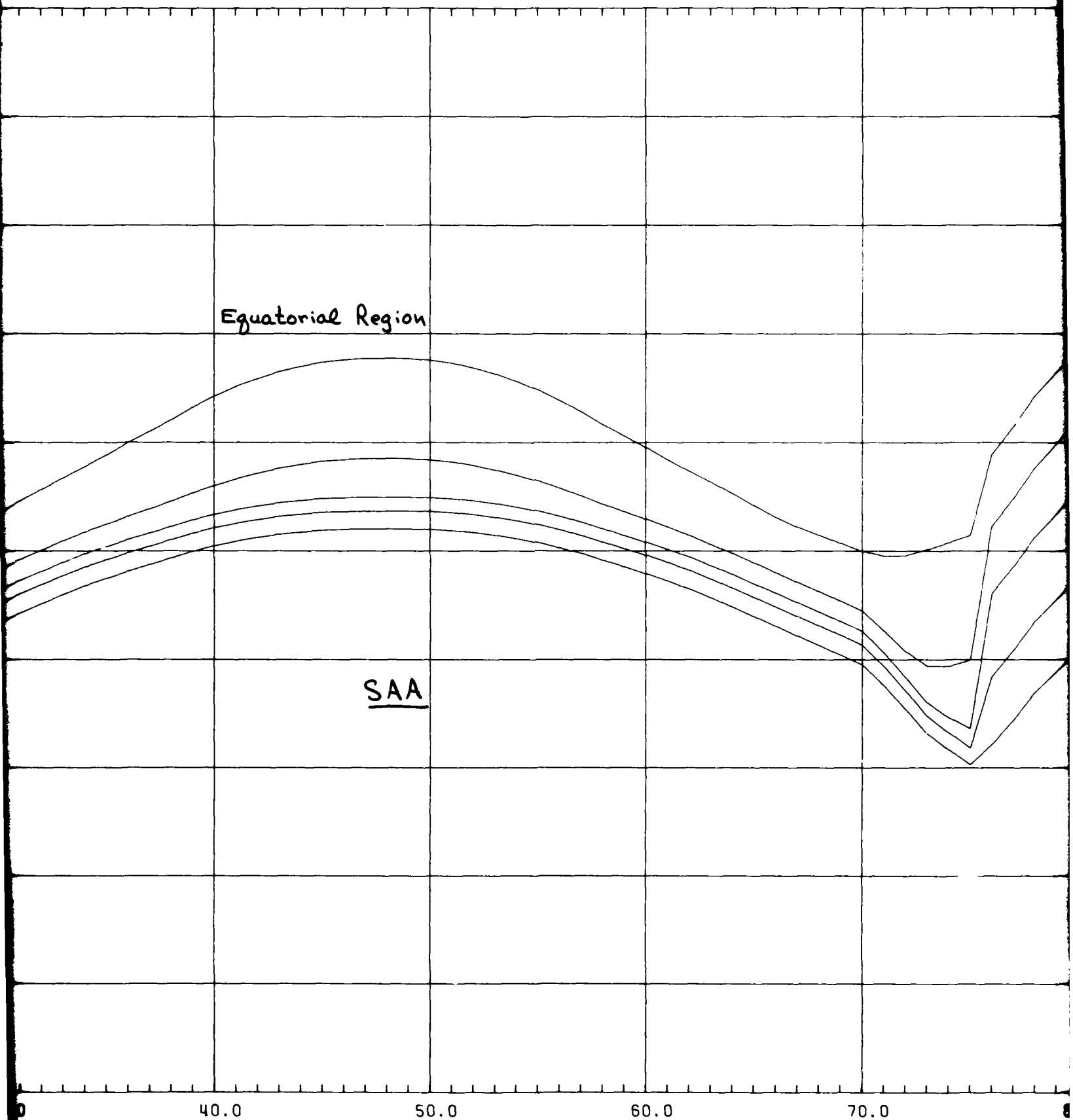
START TIME ON TAPR = 4.299995

E. G. STASSINOPOULOS-1981

5

Equatorial Region

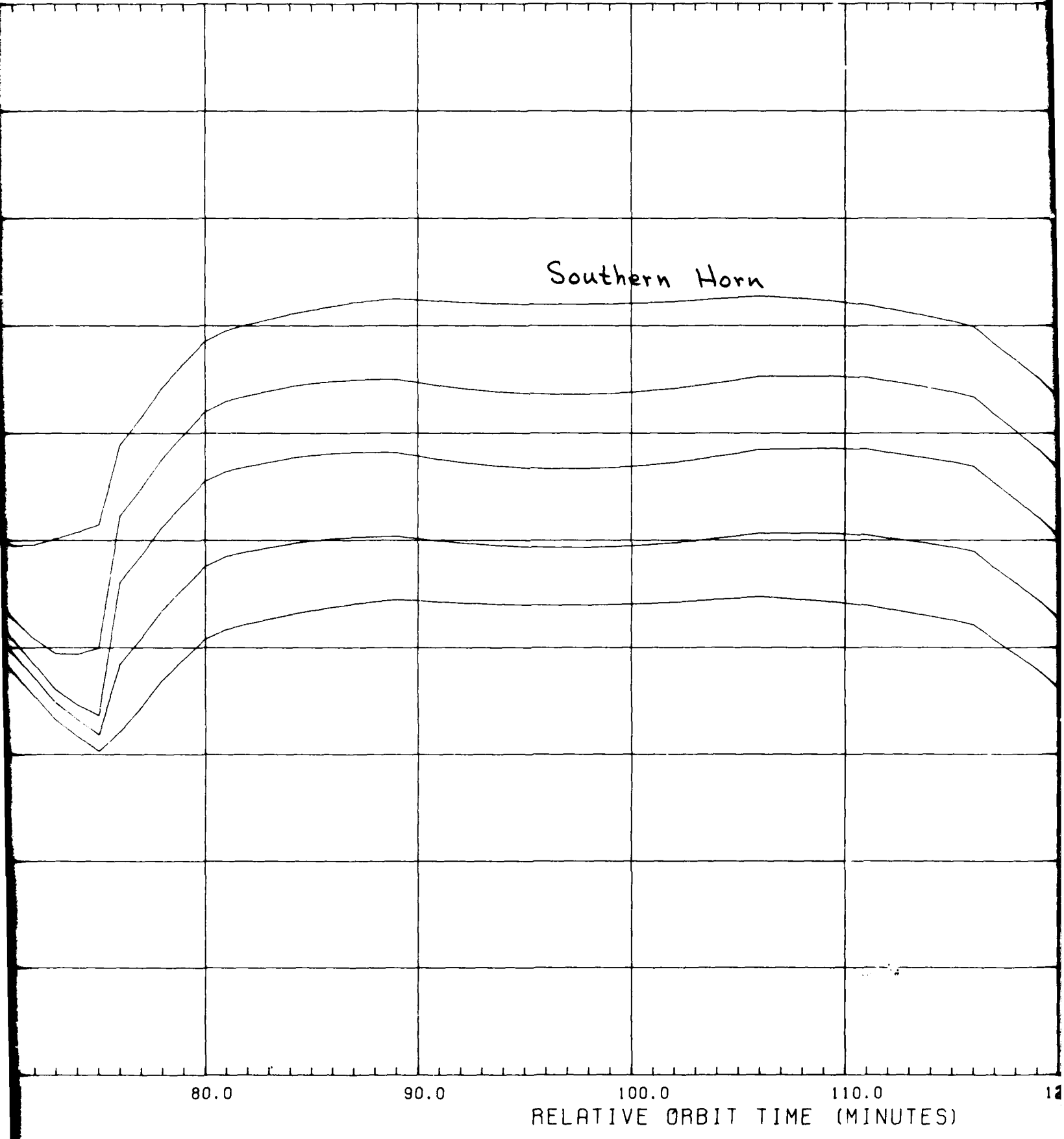
SAA



1  
2

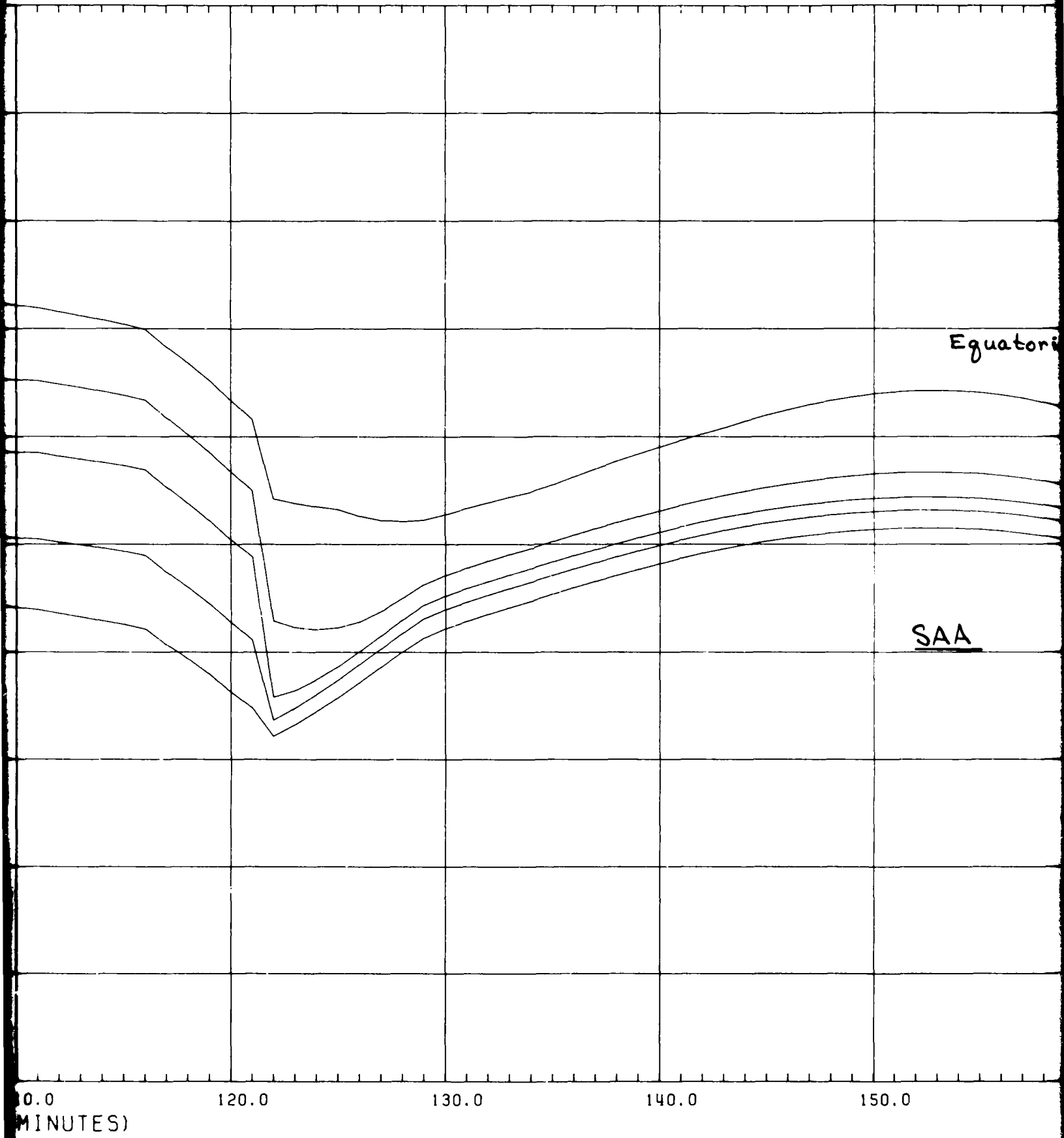
DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINUM

Southern Horn



14

# FINITE ALUMINUM SLAB SHIELDS



Equator

SAA

0.0  
MINUTES)

120.0

130.0

140.0

150.0

5

ORBIT: NA  
60 DGR/

EPOCH: 19

MODELS:  
FIELD: B  
TRAPPED  
INNER ZN  
OUTER ZN  
MISSION D  
EVALUATION

UN FACTORS

Equatorial Region

Northern  
Horn

SAA

STOP TIME

150.0 160.0 170.0 180.0 190.0

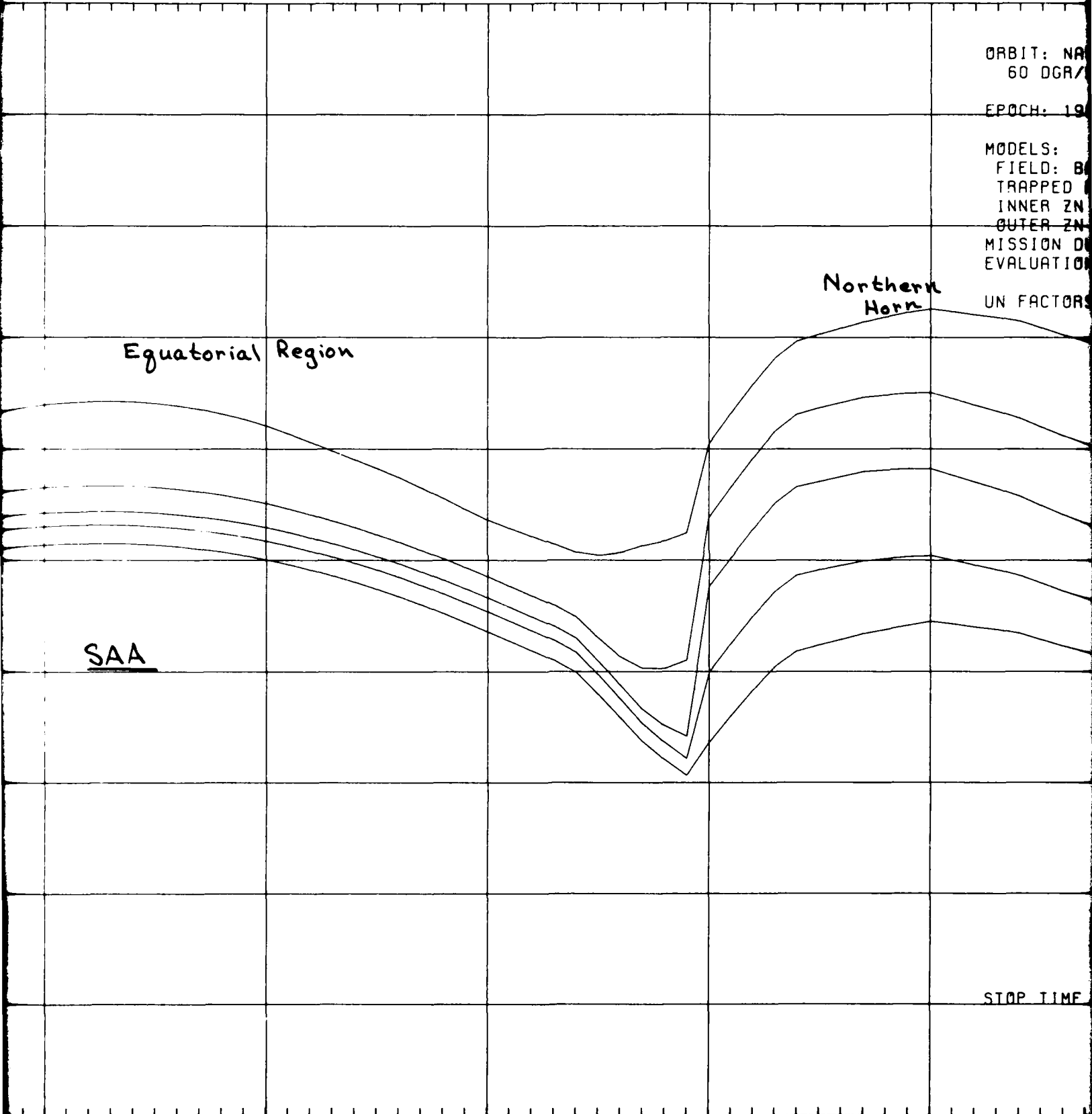




Figure 140

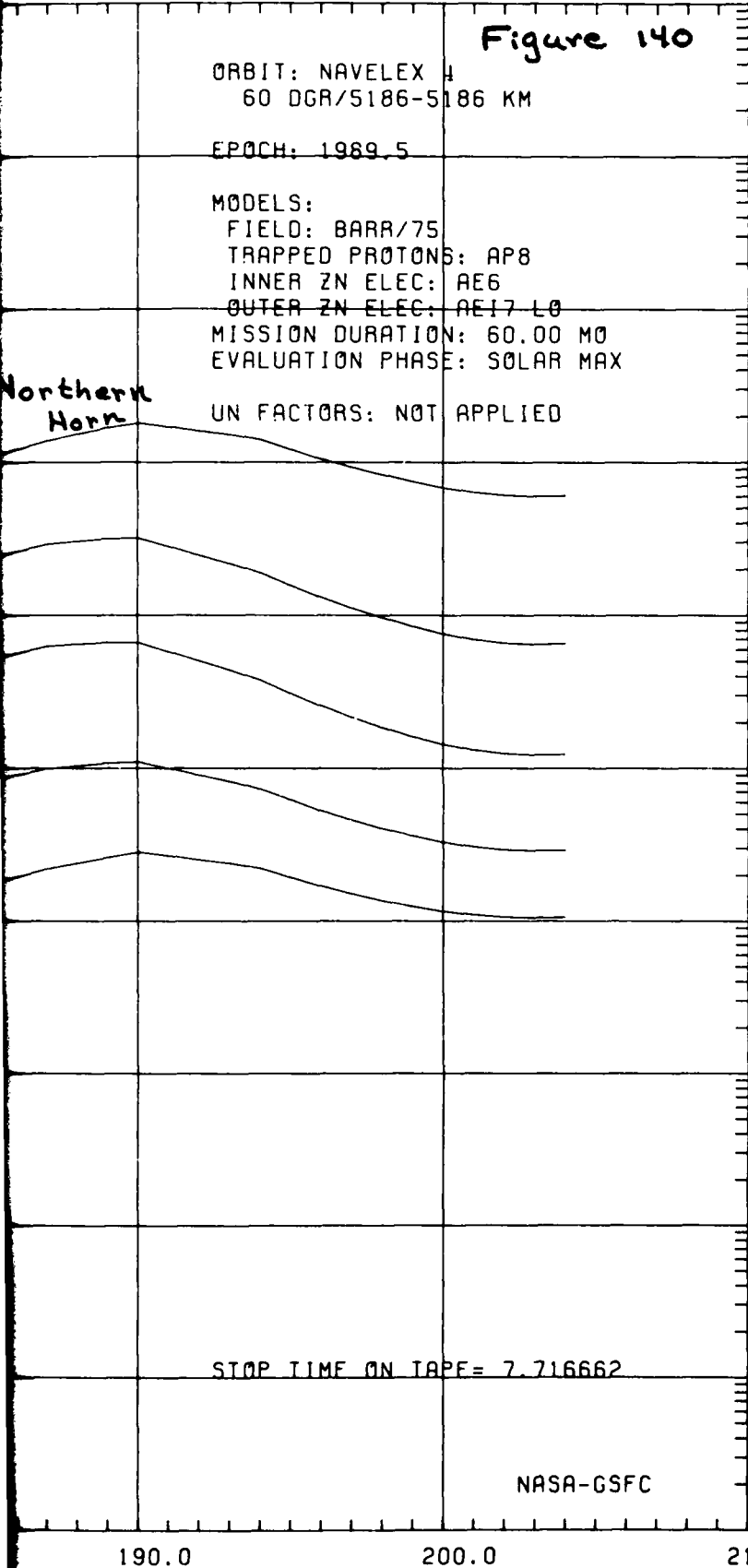
ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn



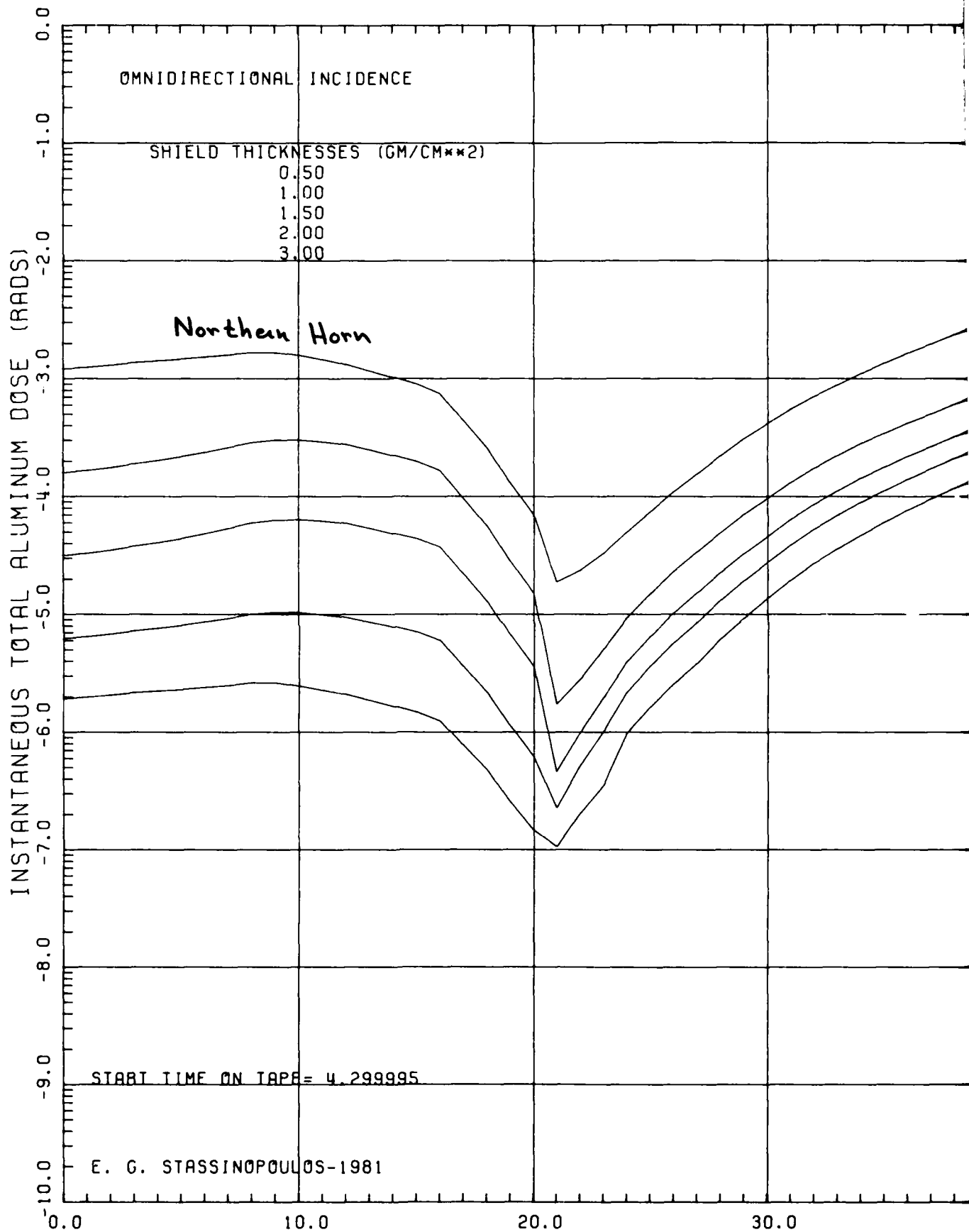
STOP TIME ON TAPE = 7.716662

NASA-GSFC

190.0

200.0

210.0



1  
2

Equatorial Region

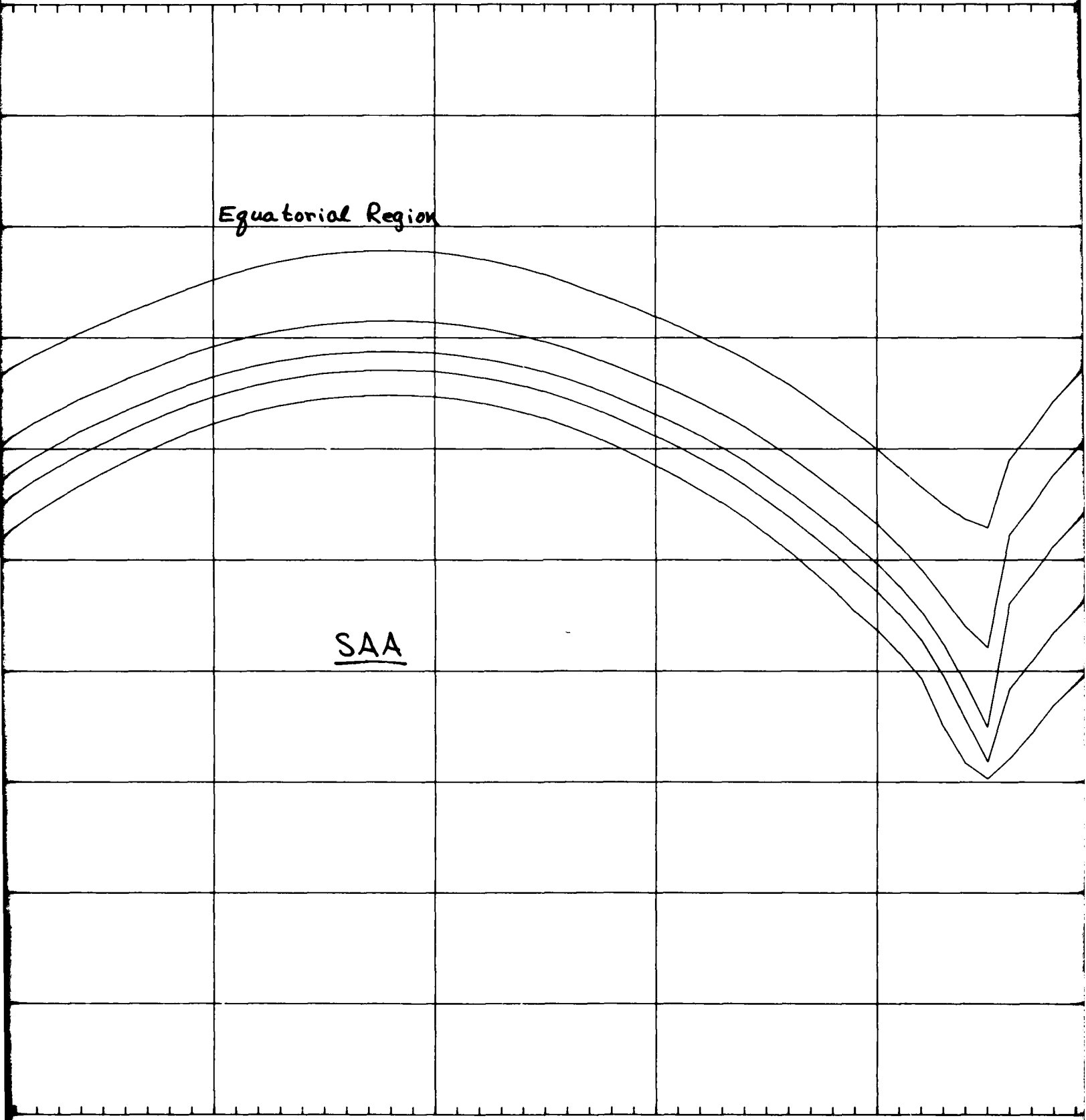
SAA

40.0

50.0

60.0

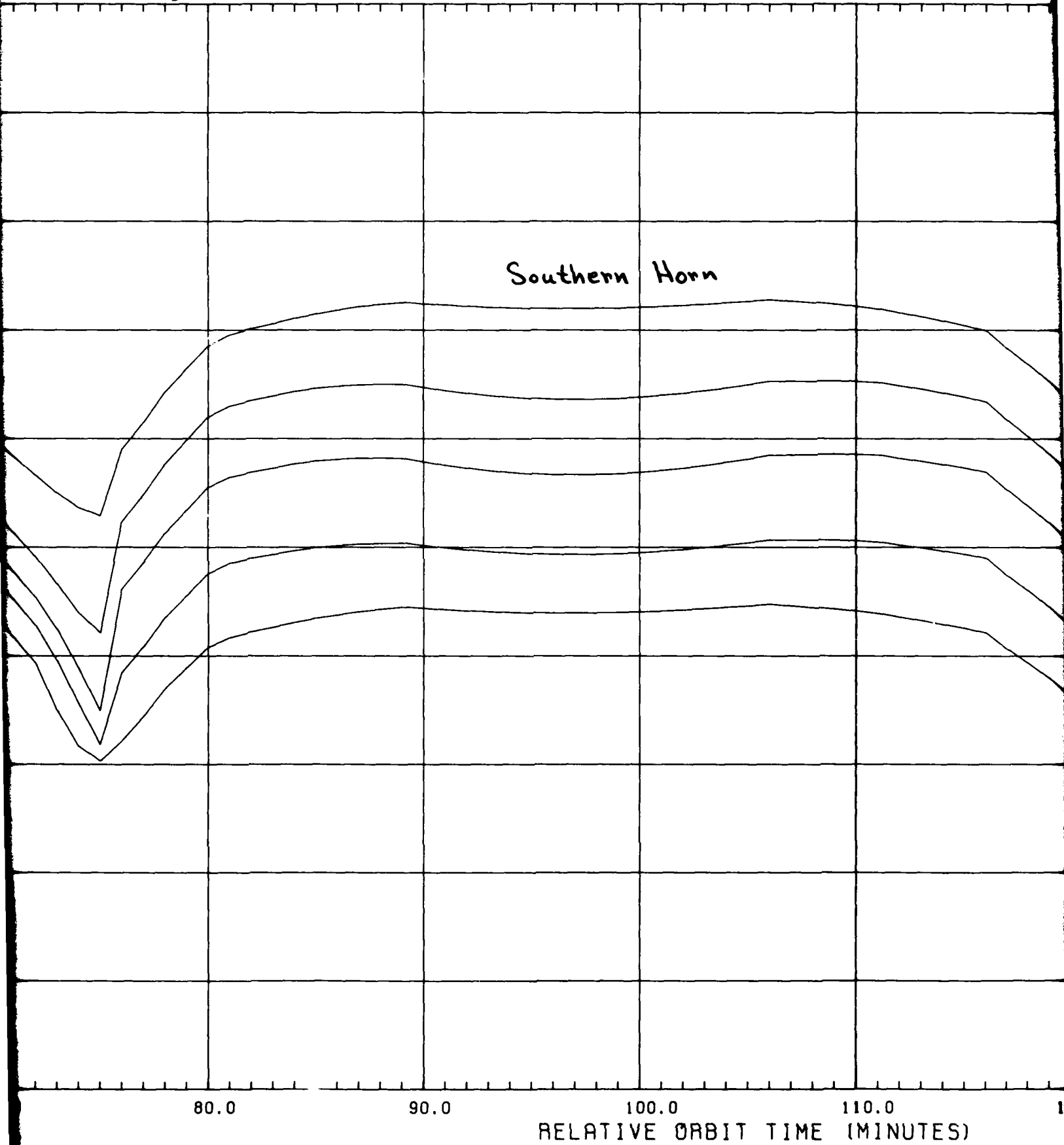
70.0



3'

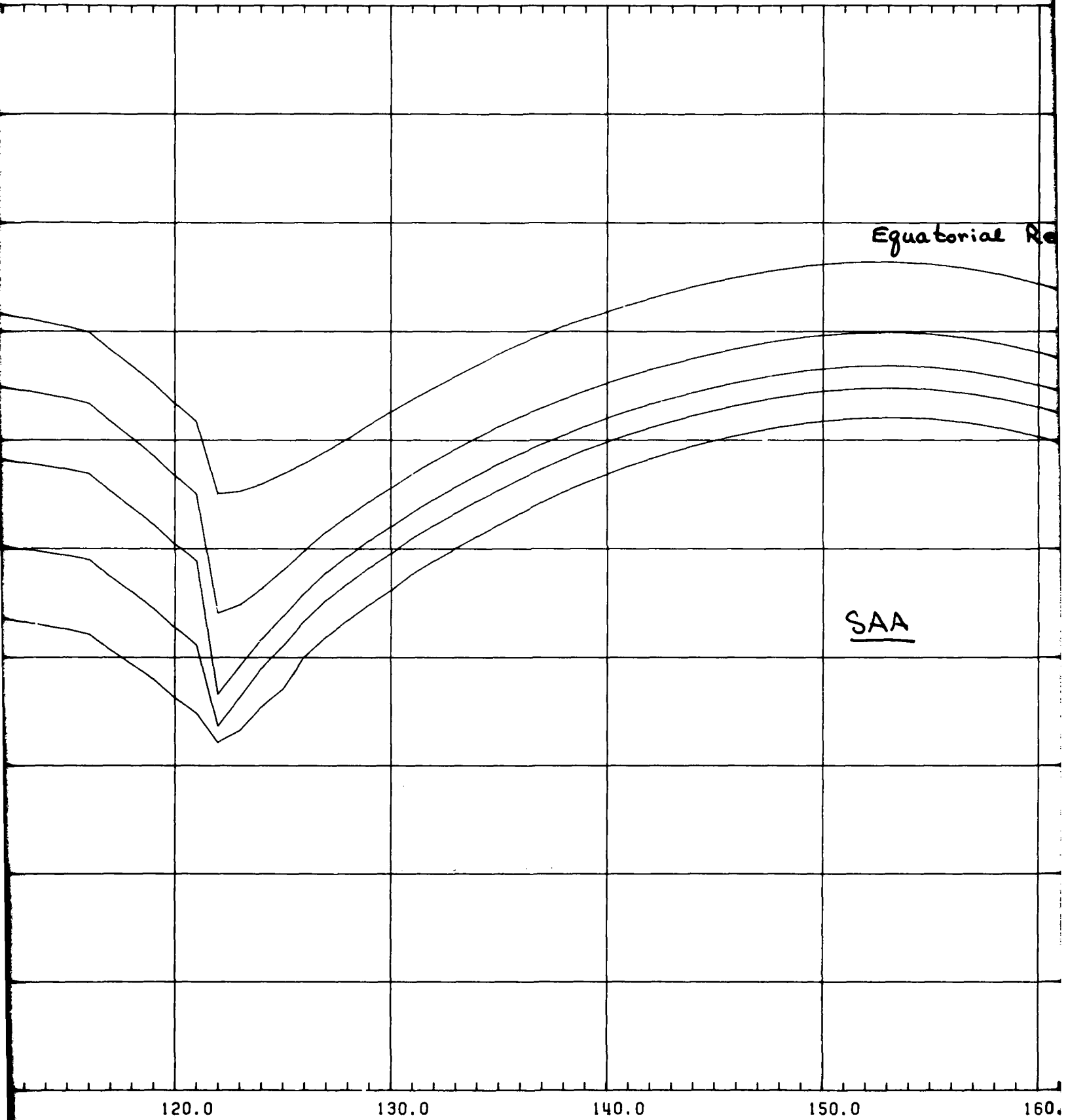
DOSE AT TRANSMISSION SURFACE OF FINITE ALUMINU

Southern Horn



14

TE ALUMINUM SLAB SHIELDS



Equatorial Re

SAA

120.0

130.0

140.0

150.0

160.0

TES)

5

ORBIT: NAVELEX  
60 DGR/5186-

EPOCH: 1989.5

MODELS:  
FIELD: BARR/7  
TRAPPED PROTONS  
INNER ZN ELECTRON  
OUTER ZN ELECTRON  
MISSION DURATION  
EVALUATION PHASE

UN FACTORS: NONE

Equatorial Region

Northern  
Horn

SAA

STOP TIME ON T

0 160.0 170.0 180.0 190.0

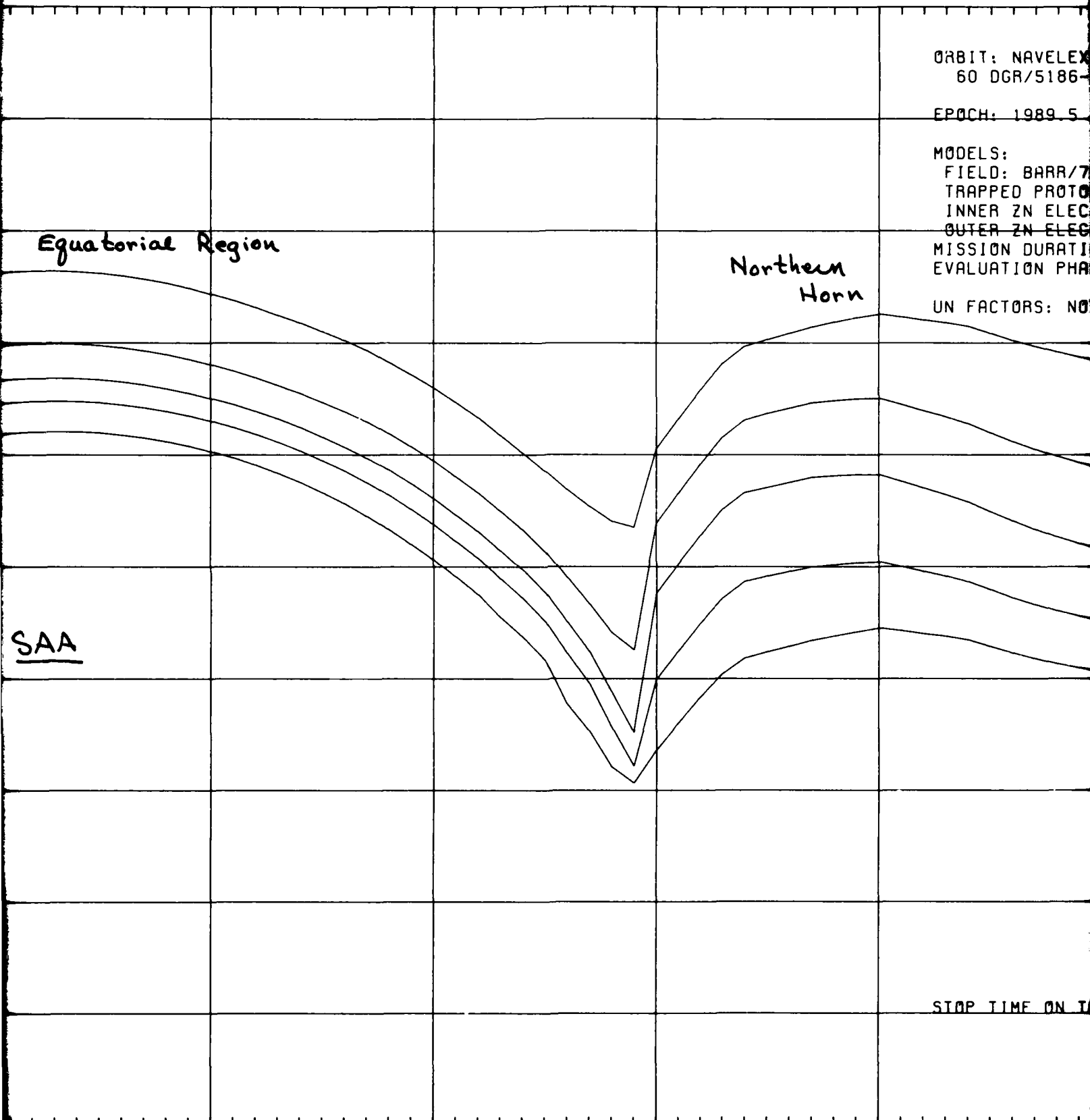


Figure 141

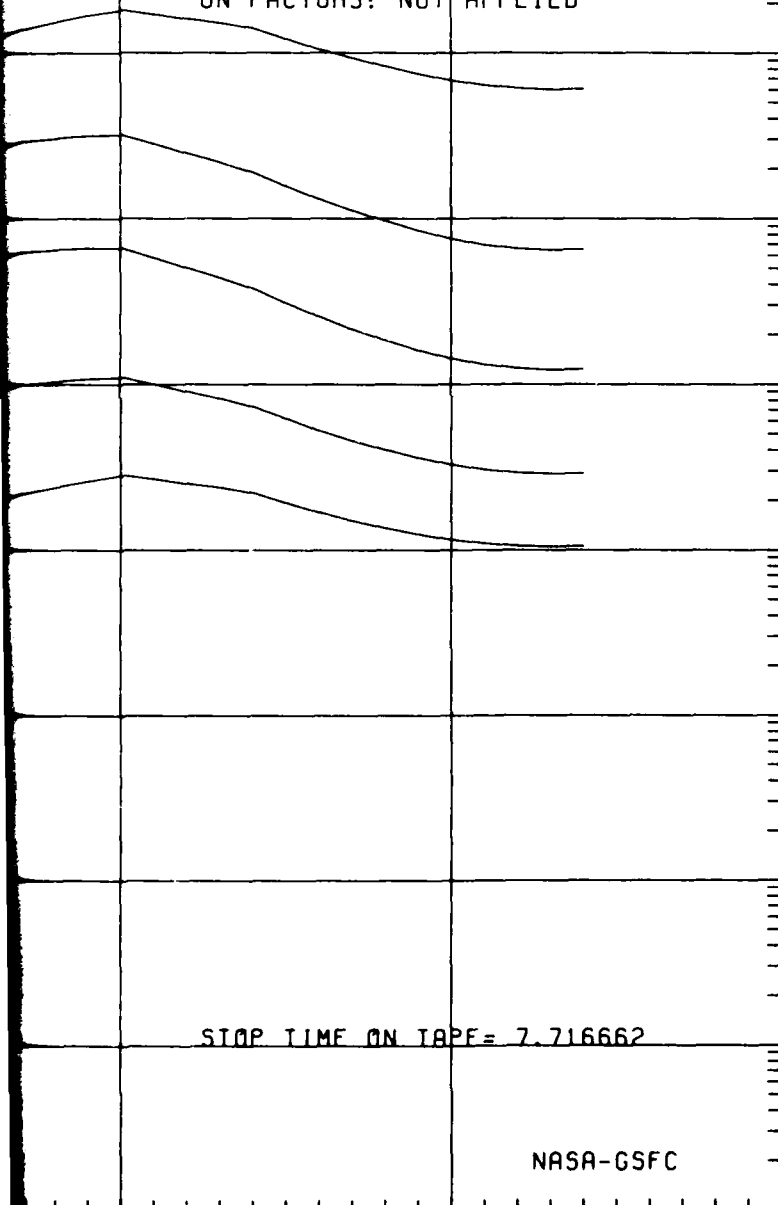
ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 M0  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

sun  
Horn



STOP TIME ON TAPE = 7.716662

NASA-GSFC

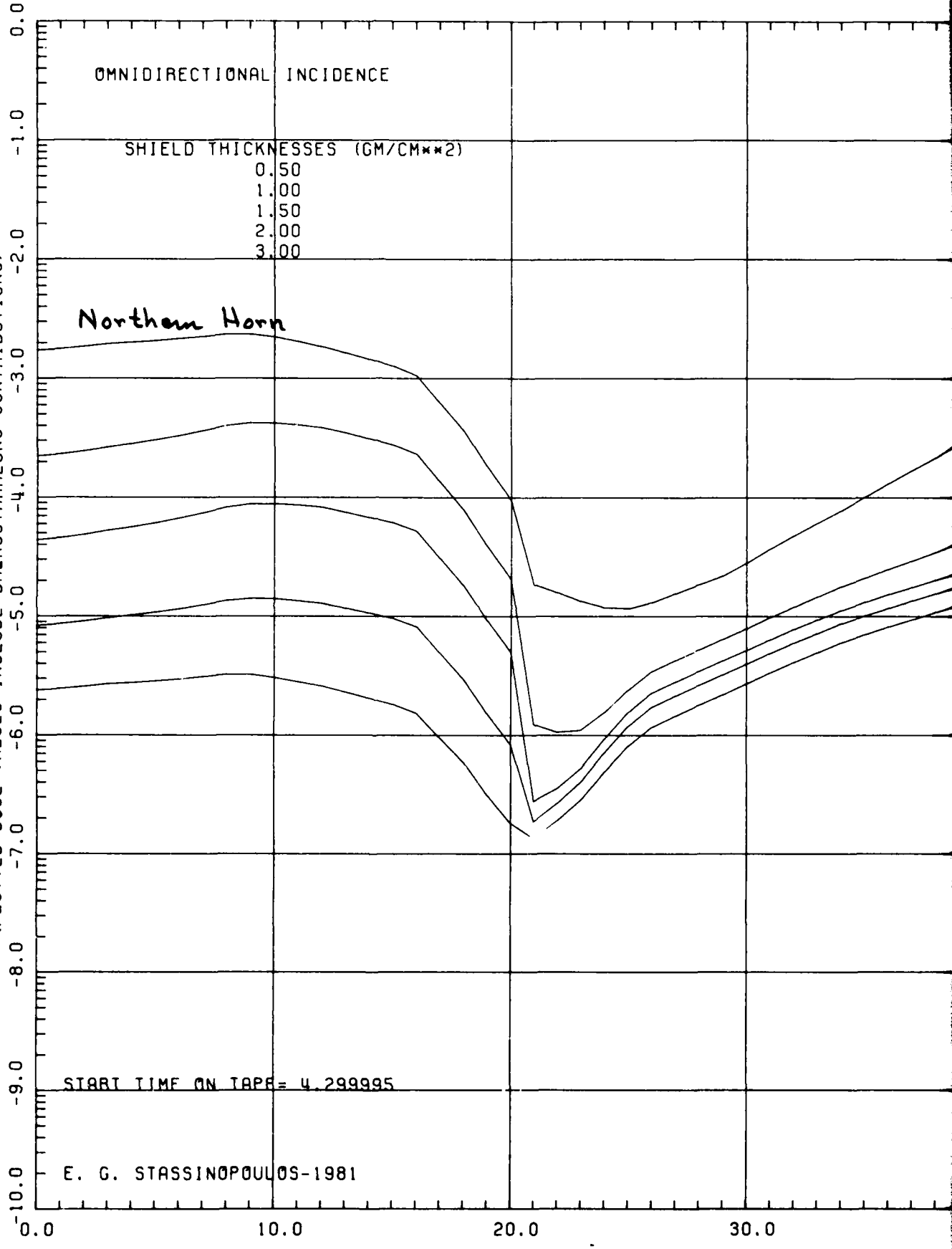
190.0

200.0

210.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)





2

Equatorial Region

SAA

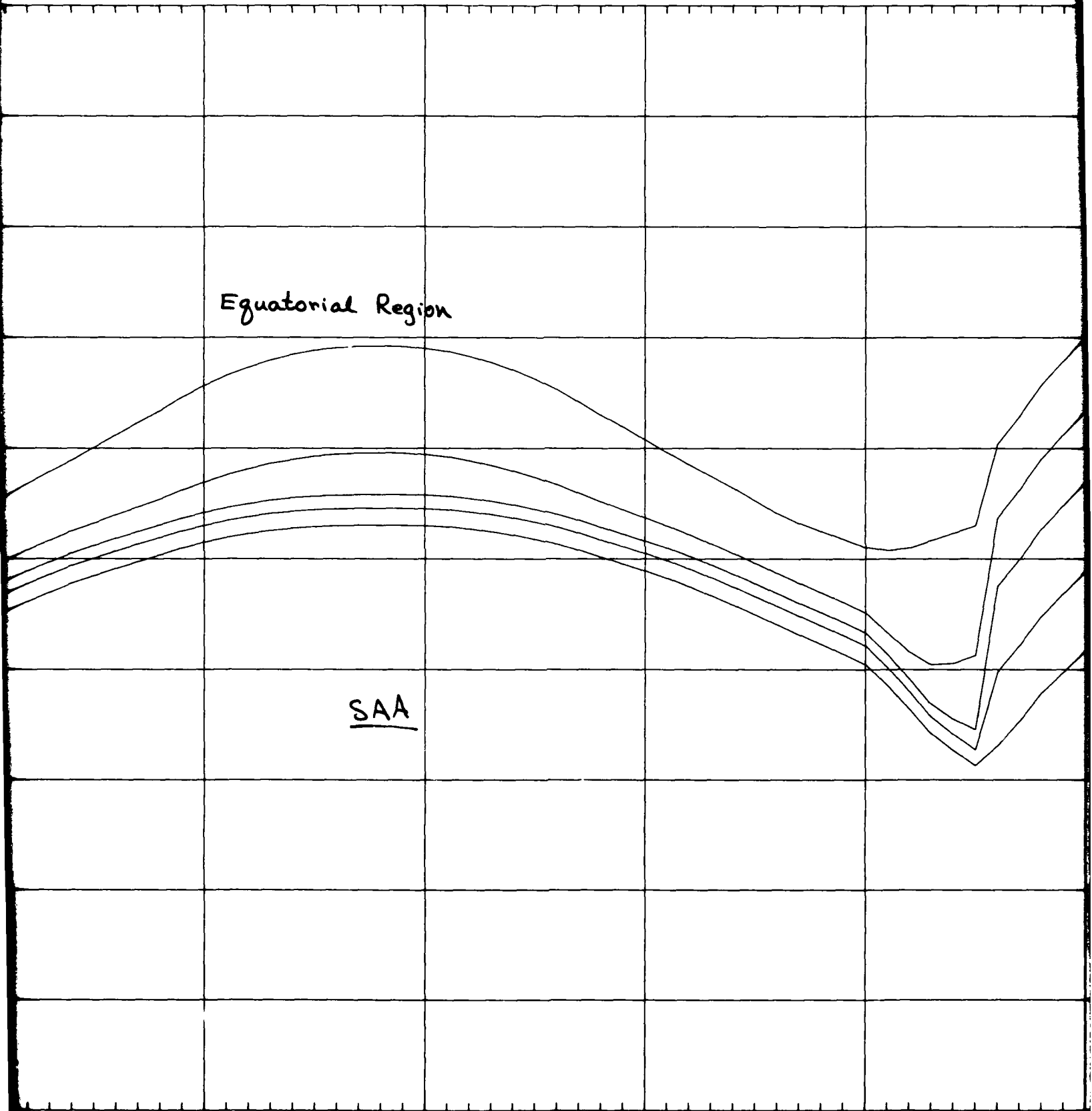
40.0

50.0

60.0

70.0

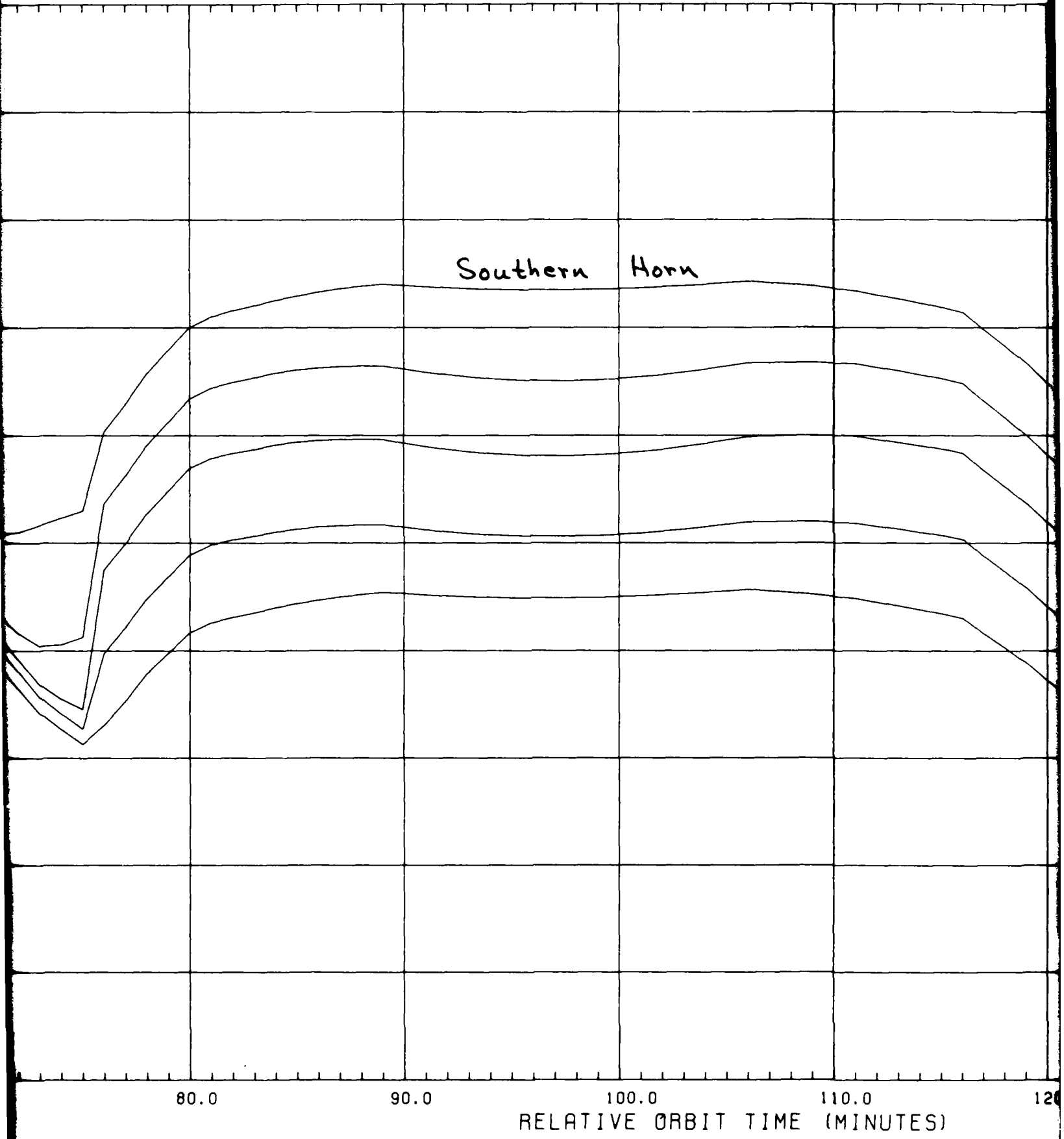
80.0



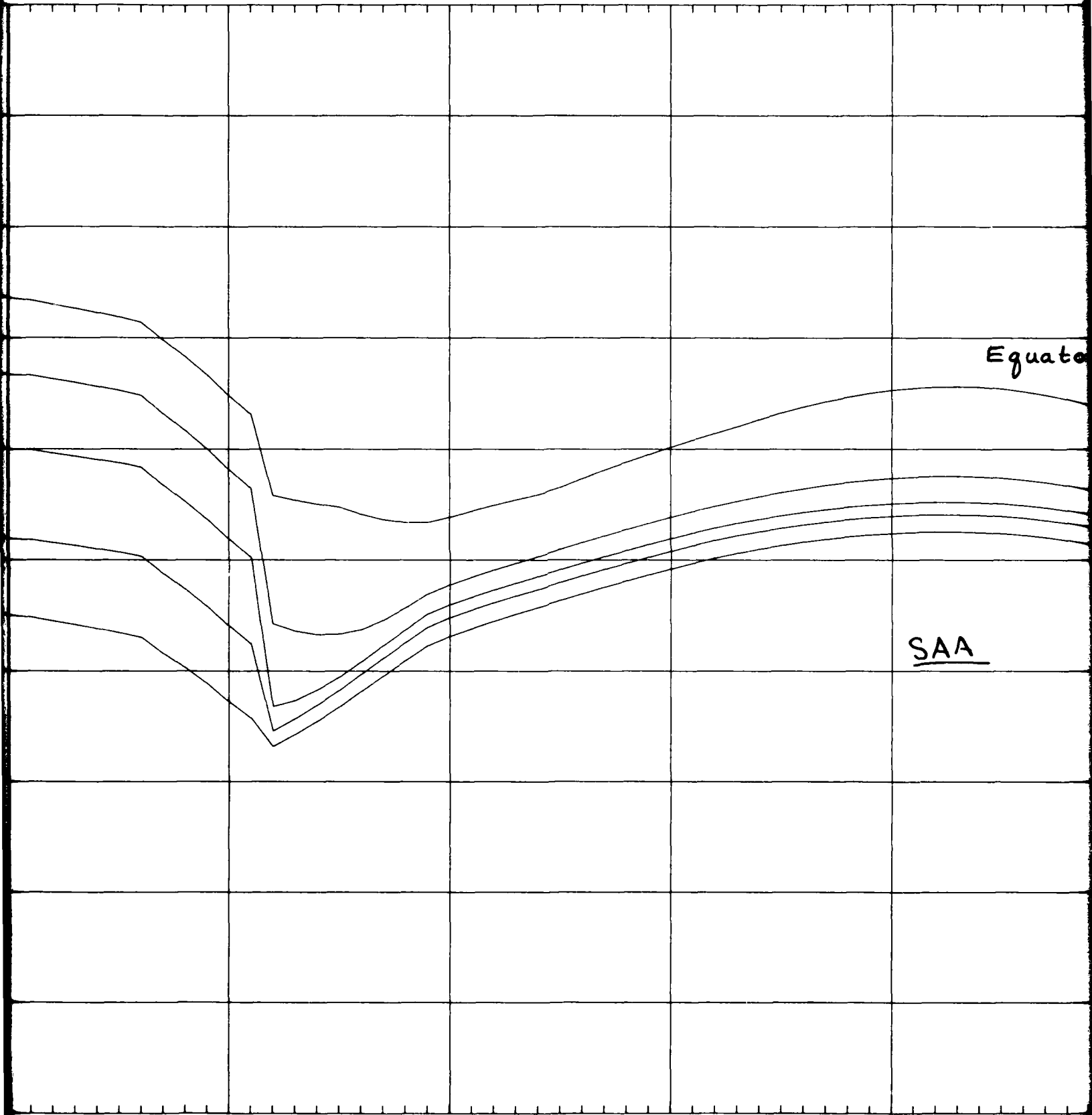
3

DOSE IN SEMI-INFINITE ALUMINUM MEDIUM

Southern Horn



LUMINUM MEDIUM



Equato

SAA

0.0  
MINUTES)

120.0

130.0

140.0

150.0

ORBIT: NAVELEX  
60 DGR/5186-

EPOCH: 1989.5

MODELS:  
FIELD: BARR/7  
TRAPPED PROTONS  
INNER ZN ELECTRON  
OUTER ZN ELECTRON  
MISSION DURATION  
EVALUATION PHASE

UN FACTORS: NO

Northern  
Horn

Equatorial Region

SAA

STOP TIME ON T

160.0 170.0 180.0 190.0

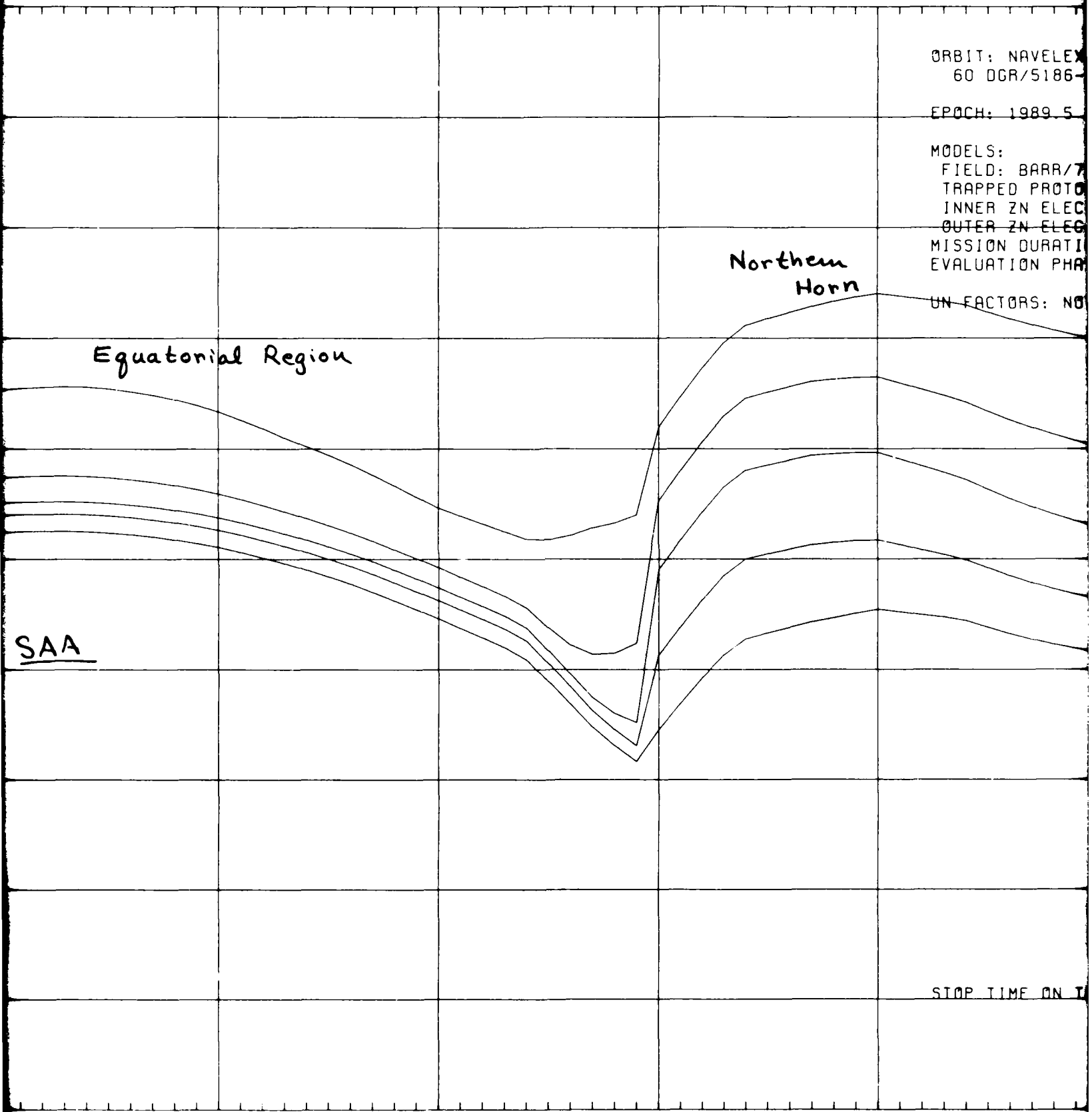


Figure 142

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AE6

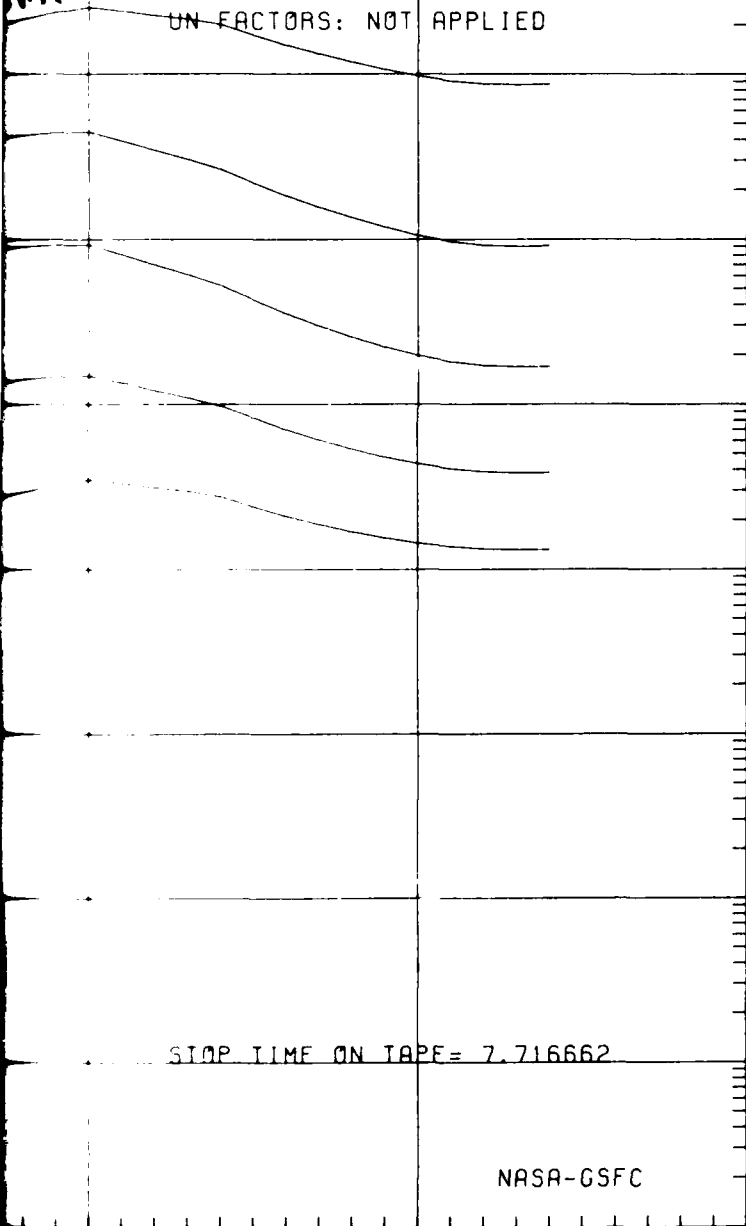
OUTER ZN ELEC: AE17-L0

MISSION DURATION: 60.00 MO

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

m  
rn



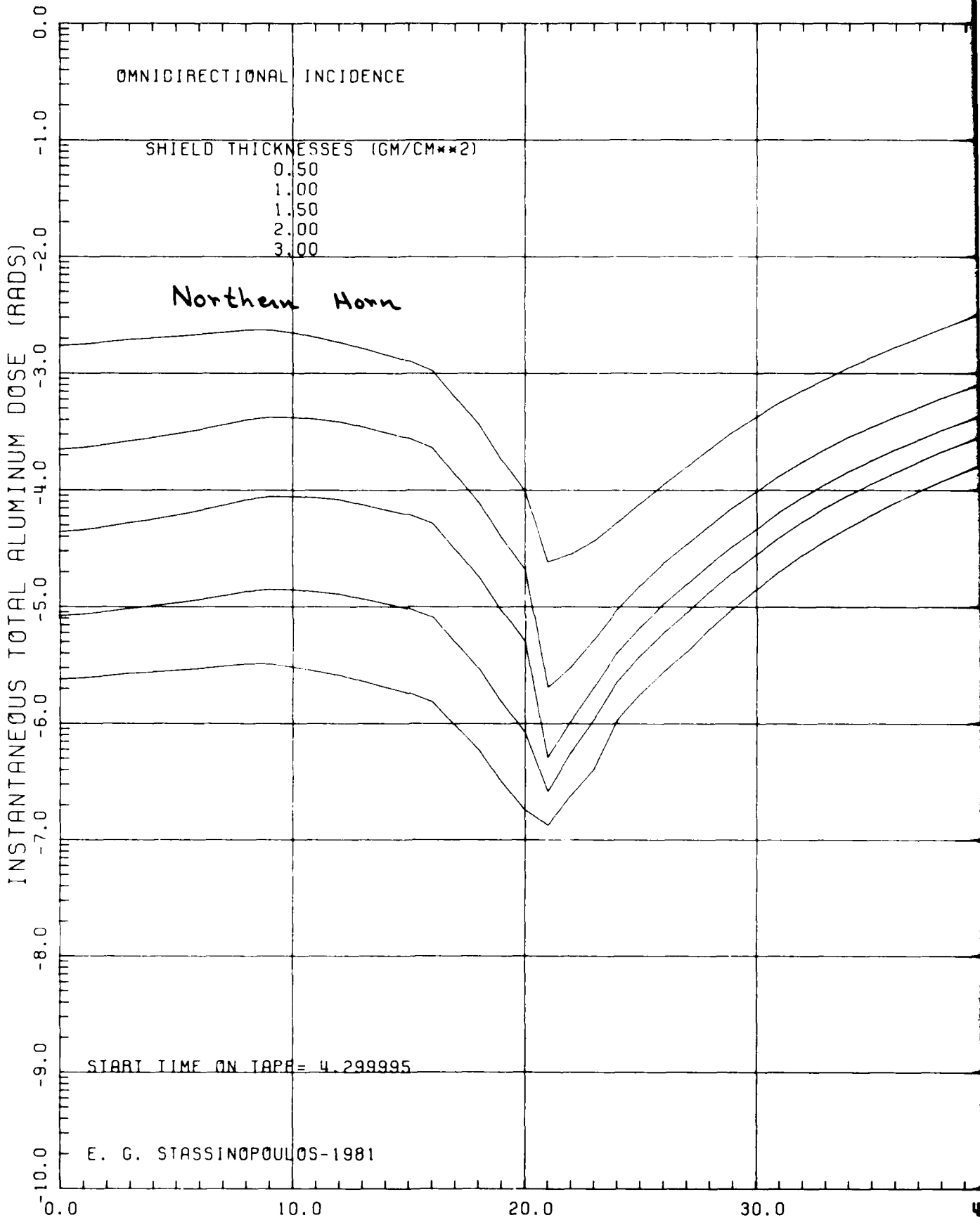
STOP TIME ON TAPE = 7.716662

NASA-GSFC

190.0

200.0

210.0



Equatorial Region

SAA

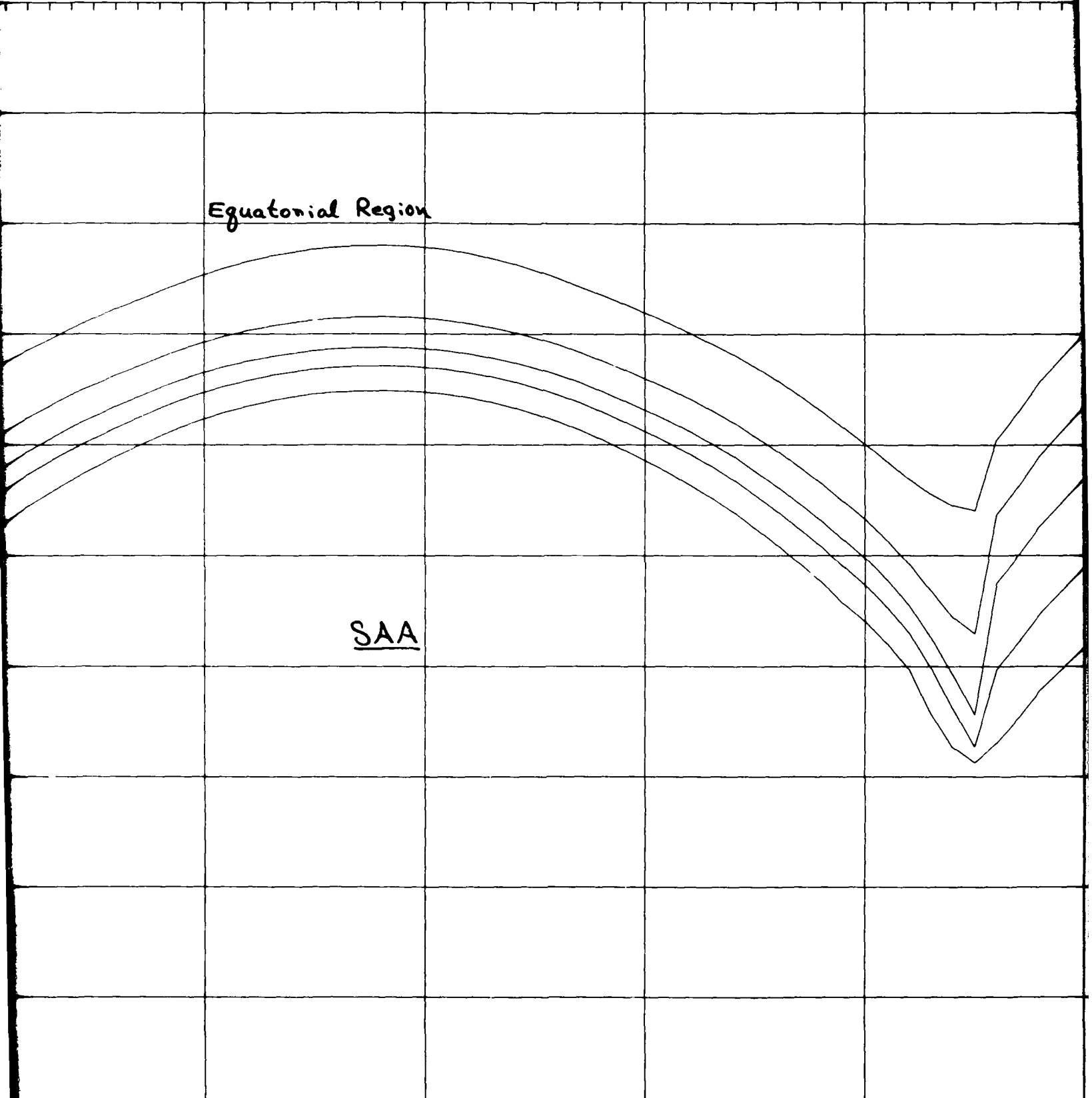
40.0

50.0

60.0

70.0

80



DOSE IN SEMI-INFINITE ALUMINUM MEDIUM

Southern Horn

80.0

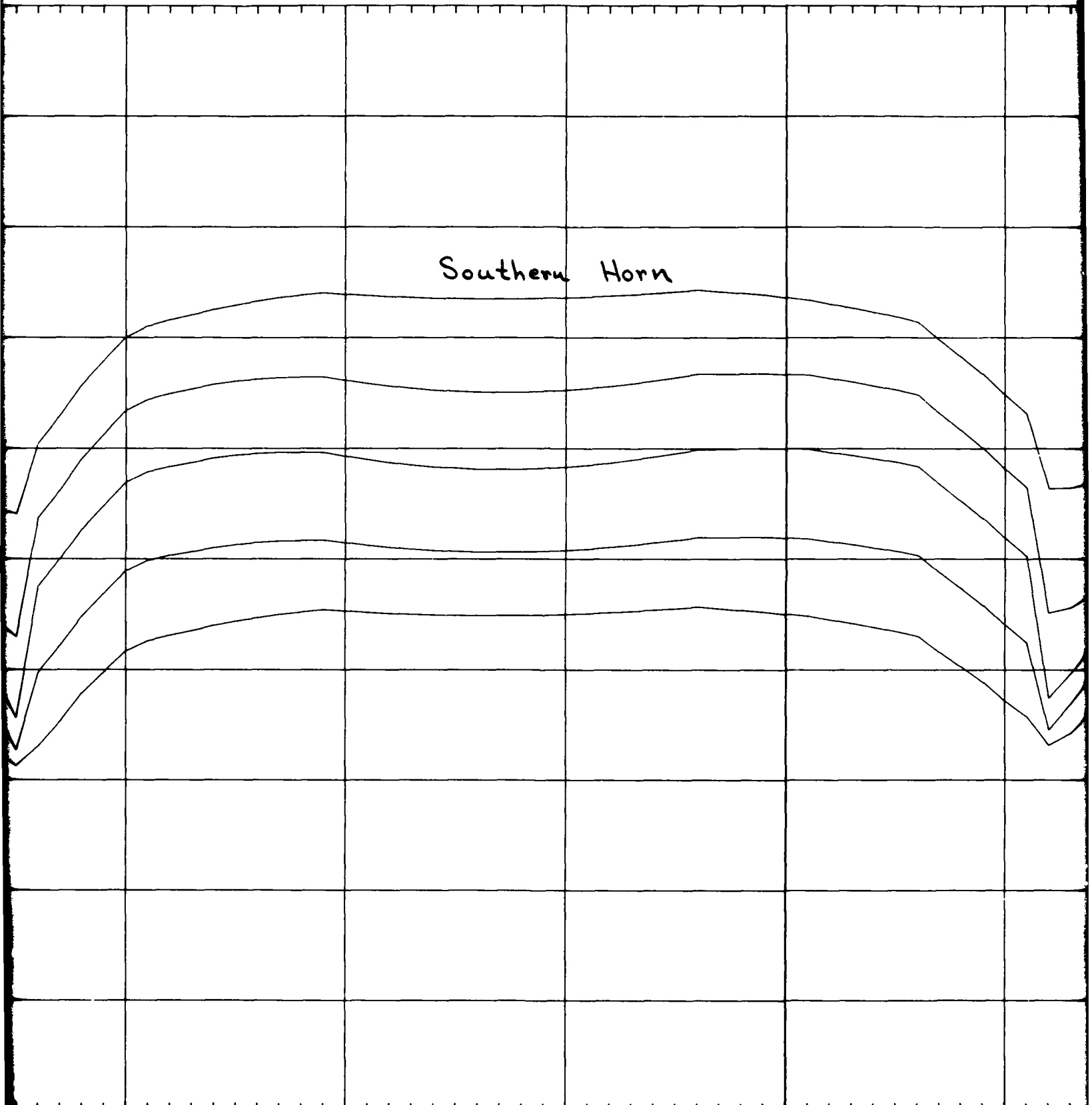
90.0

100.0

110.0

120.0

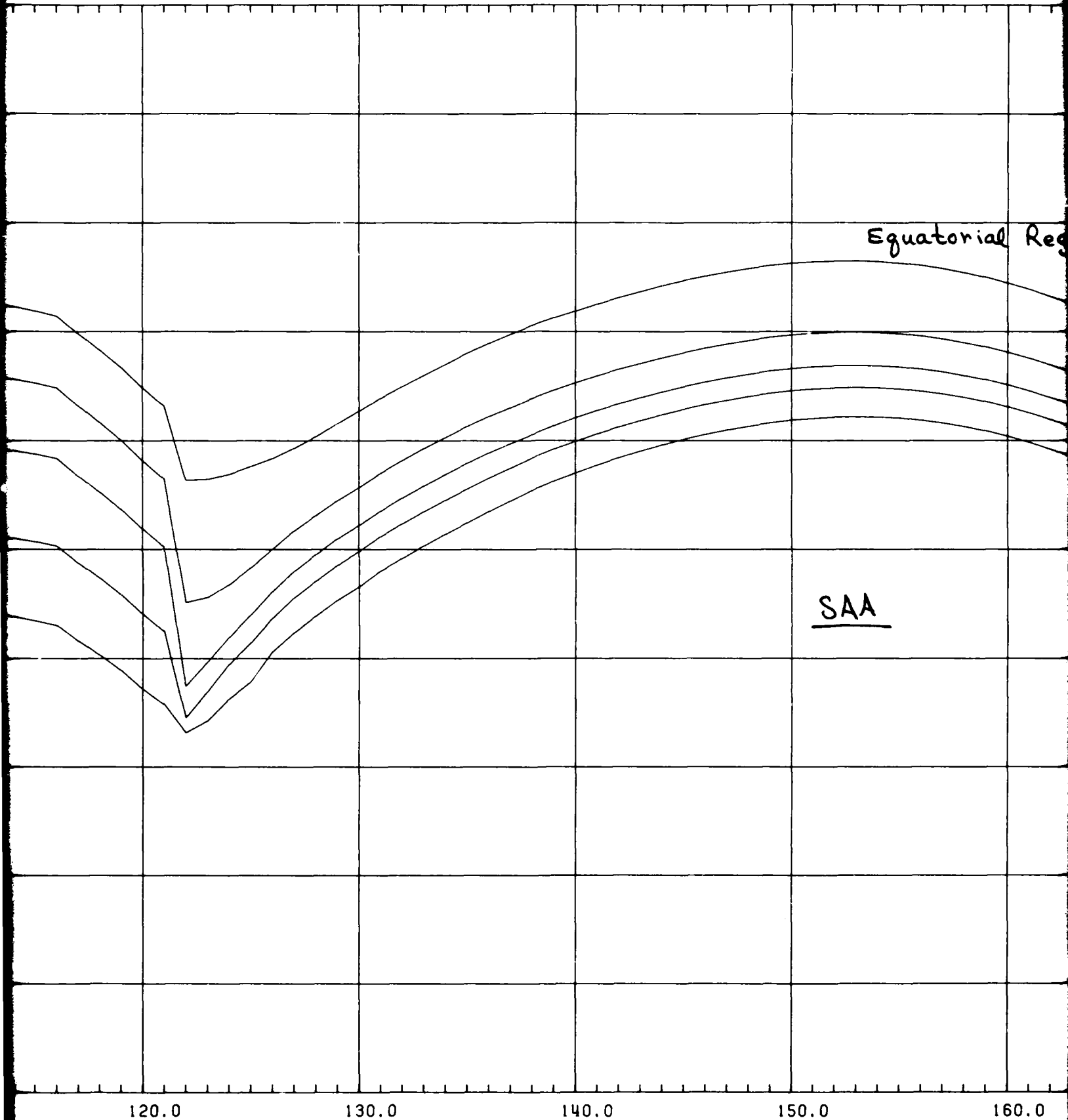
RELATIVE ORBIT TIME (MINUTES)





UM MEDIUM

4



Equatorial Reg

SAA

120.0

130.0

140.0

150.0

160.0

S)

5.

ORBIT: NAVELEX 4  
60 DGR/5186-518

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS:  
INNER ZN ELEC: A  
OUTER ZN ELEC: A  
MISSION DURATION:  
EVALUATION PHASE:  
UN FACTORS: NOT A

Equatorial Region

Northern  
Horn

AA

STOP TIME ON TAPE

160.0

170.0

180.0

190.0

200.0

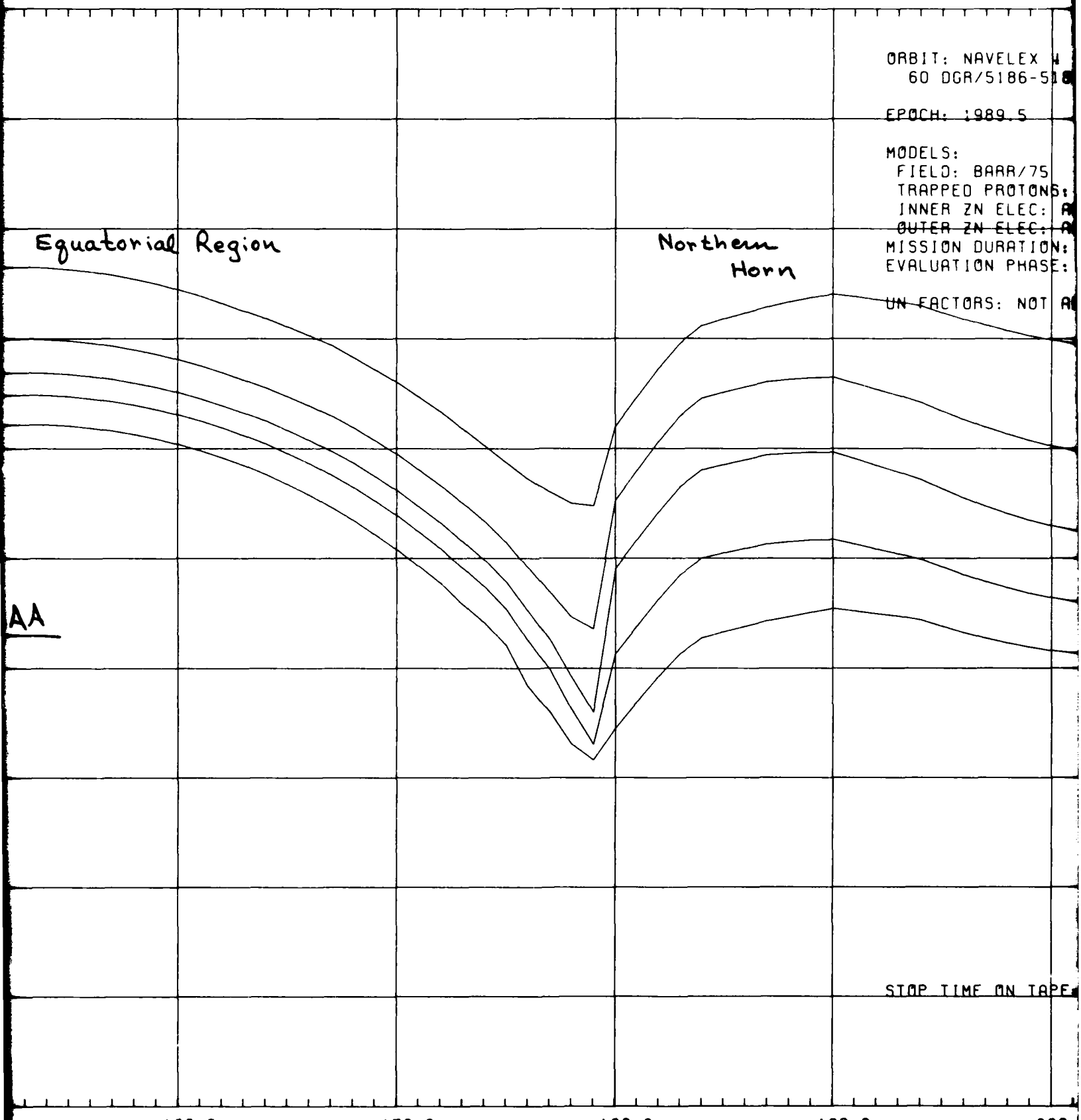


Figure 143

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 7.716662

NASA-GSFC

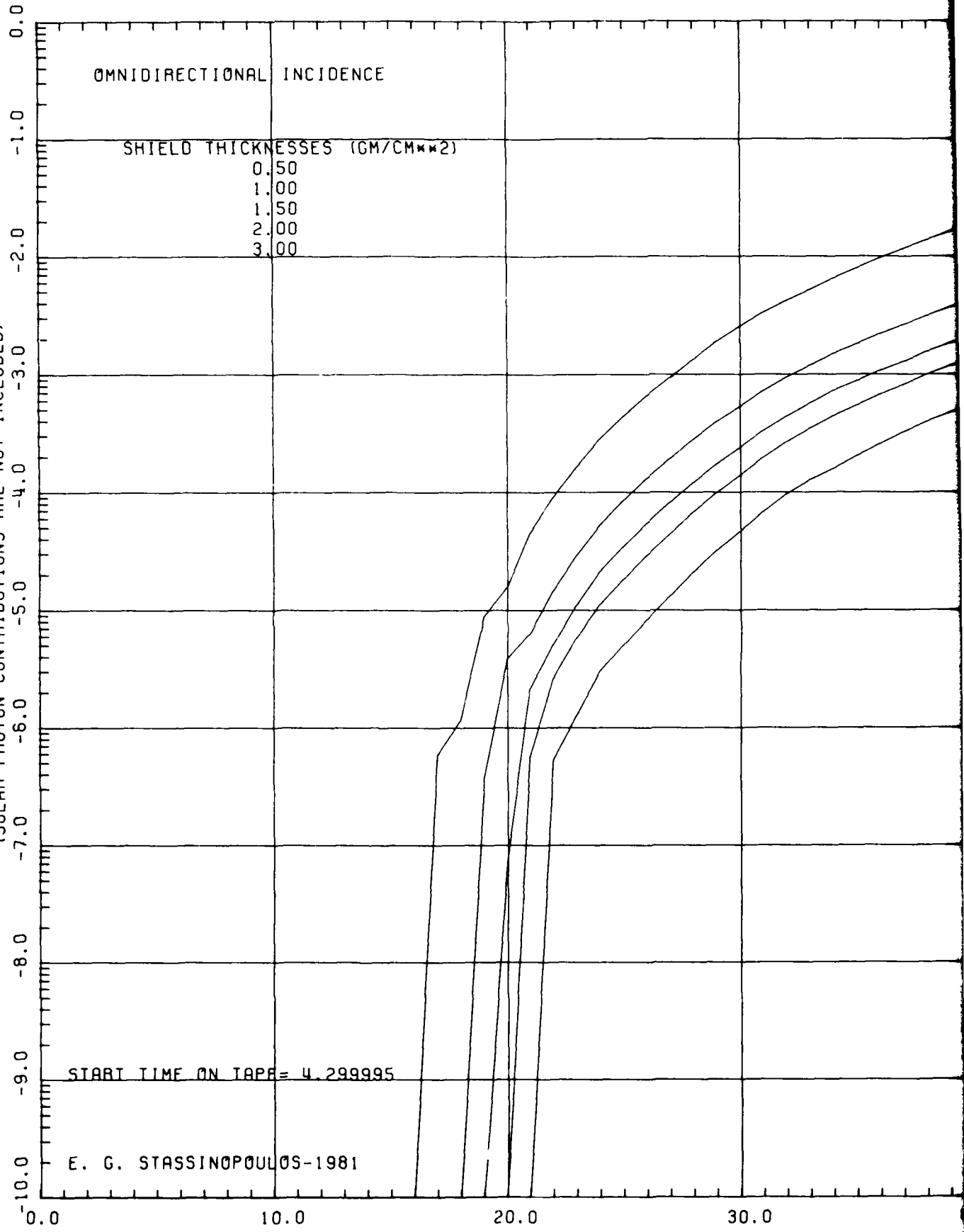
190.0

200.0

210.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



12

Equatorial Region

SAA

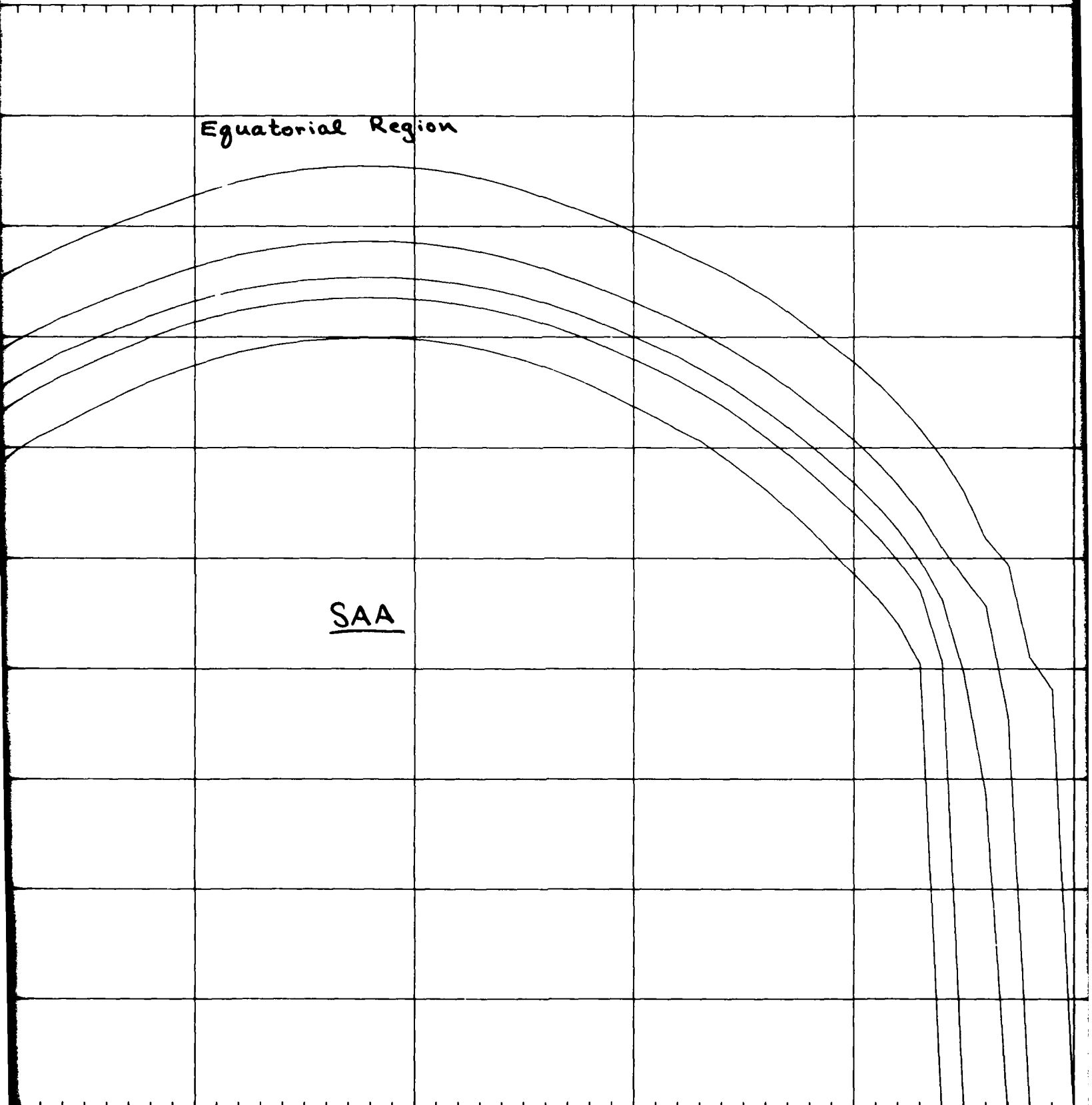
40.0

50.0

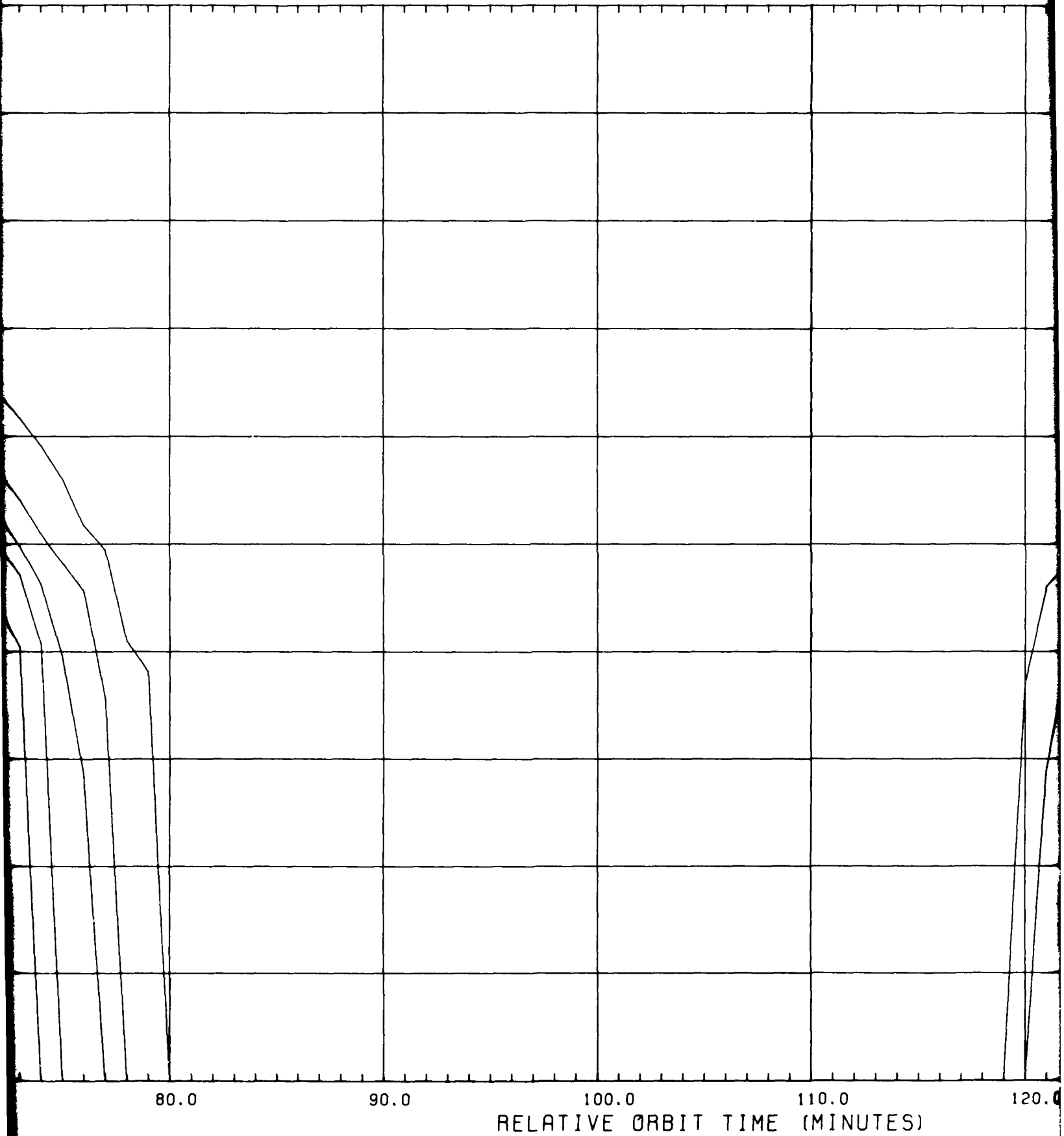
60.0

70.0

80.0



DOSE AT CENTER OF ALUMINUM SPHERES



80.0

90.0

100.0

110.0

120.0

RELATIVE ORBIT TIME (MINUTES)

ERES

4

Equatorial Region

SAA

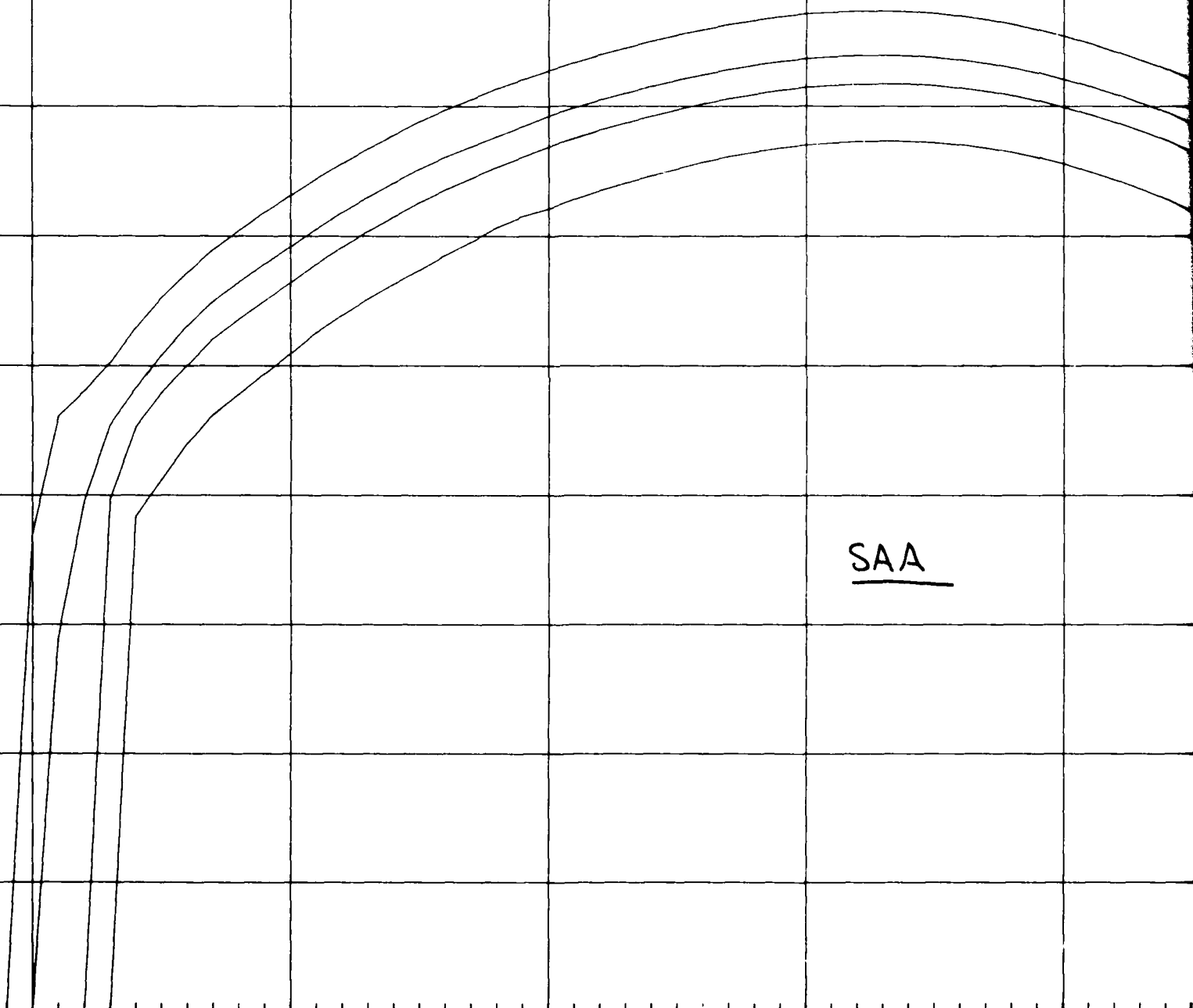
120.0

130.0

140.0

150.0

160.0



5

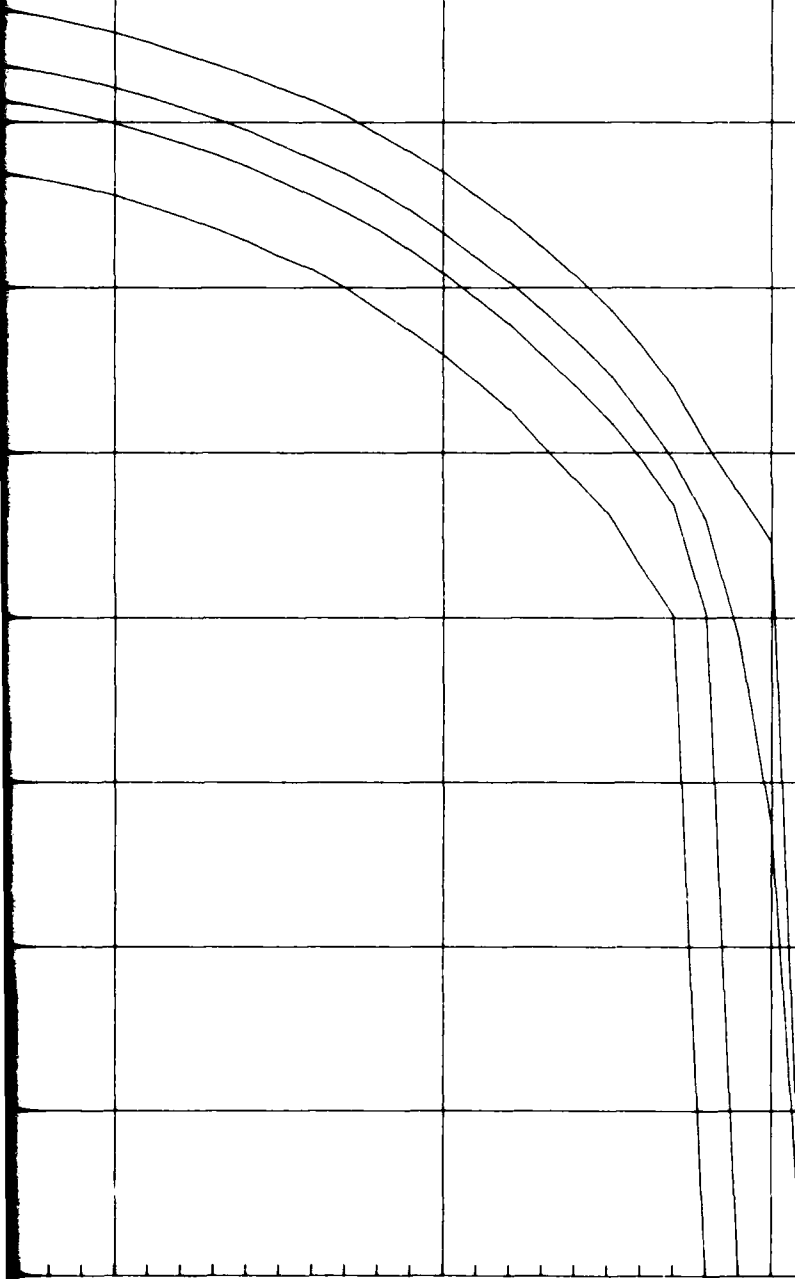
Figure

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX  
  
UN FACTORS: NOT APPLIED

Region



STOP TIME ON TAPE= 7.716662

NASA-GS

160.0                      170.0                      180.0                      190.0                      200.0



Figure 144

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:

FIELD: BARR/75

TRAPPED PROTONS: AP8

INNER ZN ELEC: AEG

OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 M0

EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 7.716662

NASA-GSFC

190.0

200.0

210.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

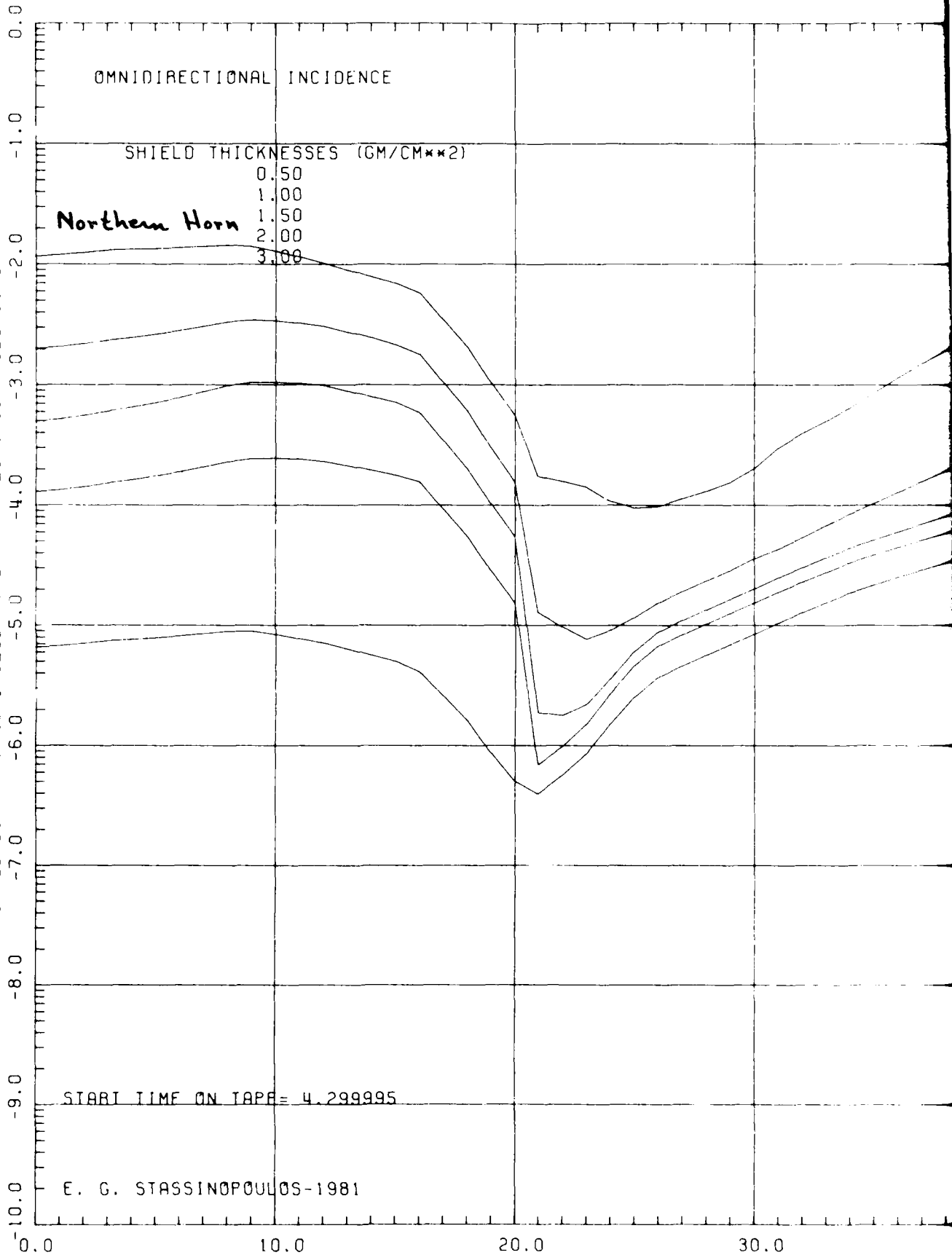
(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)

OMNIDIRECTIONAL INCIDENCE

SHIELD THICKNESSES (GM/CM\*\*2)

Northern Horn

- 0.50
- 1.00
- 1.50
- 2.00
- 3.00

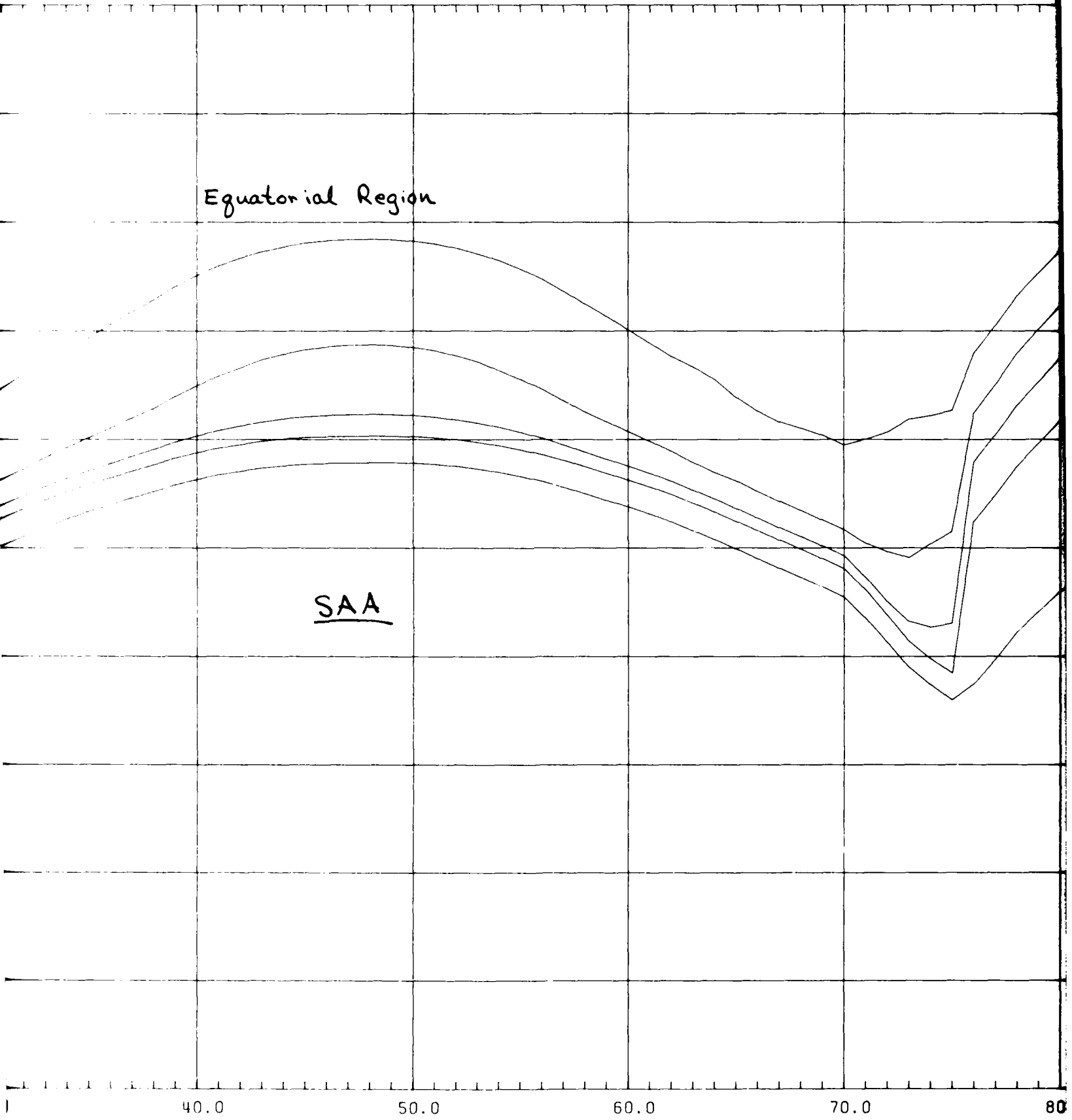


START TIME ON TAPE = 4.299995

E. G. STASSINOPOULOS-1981

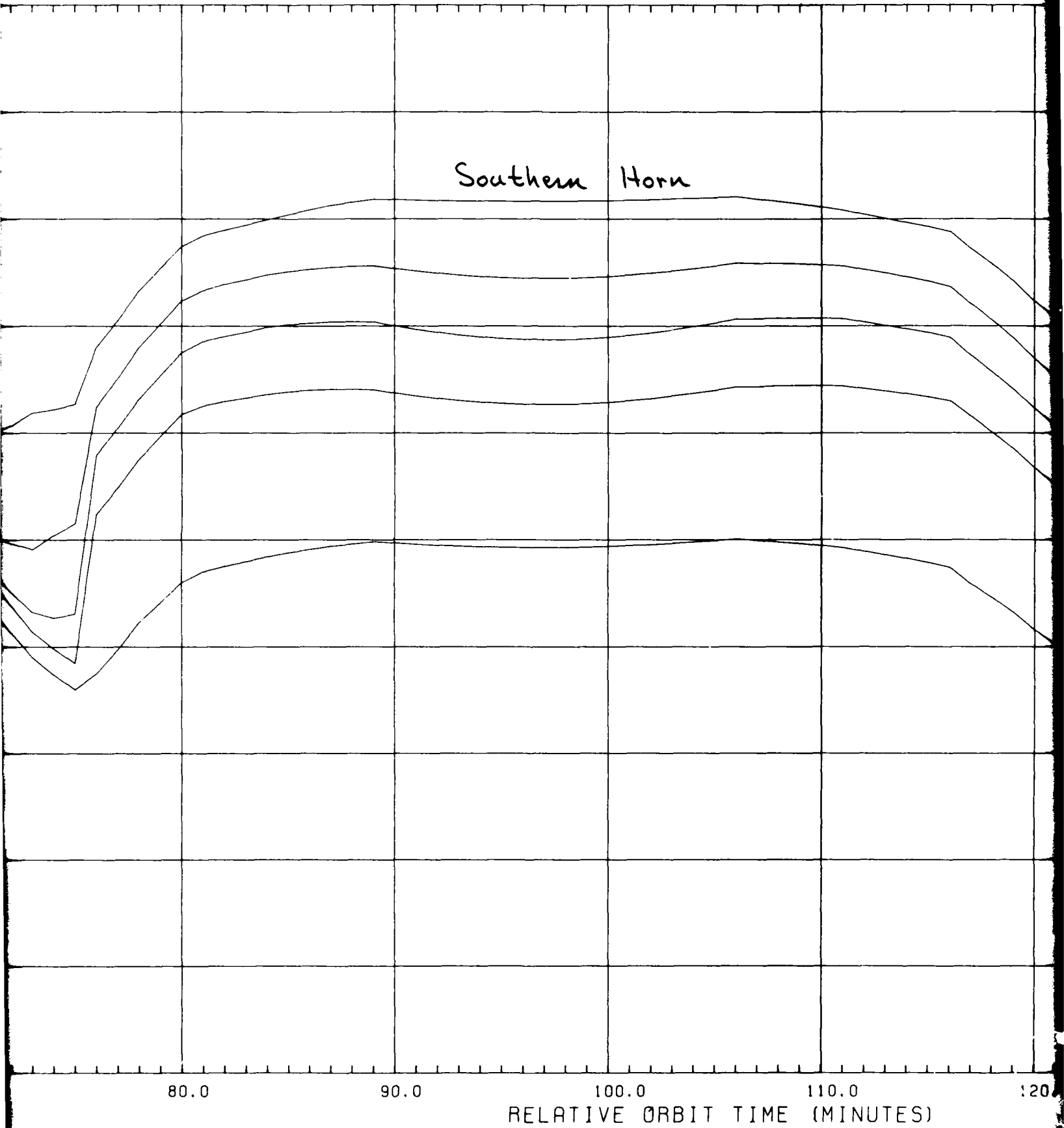
Equatorial Region

SAA



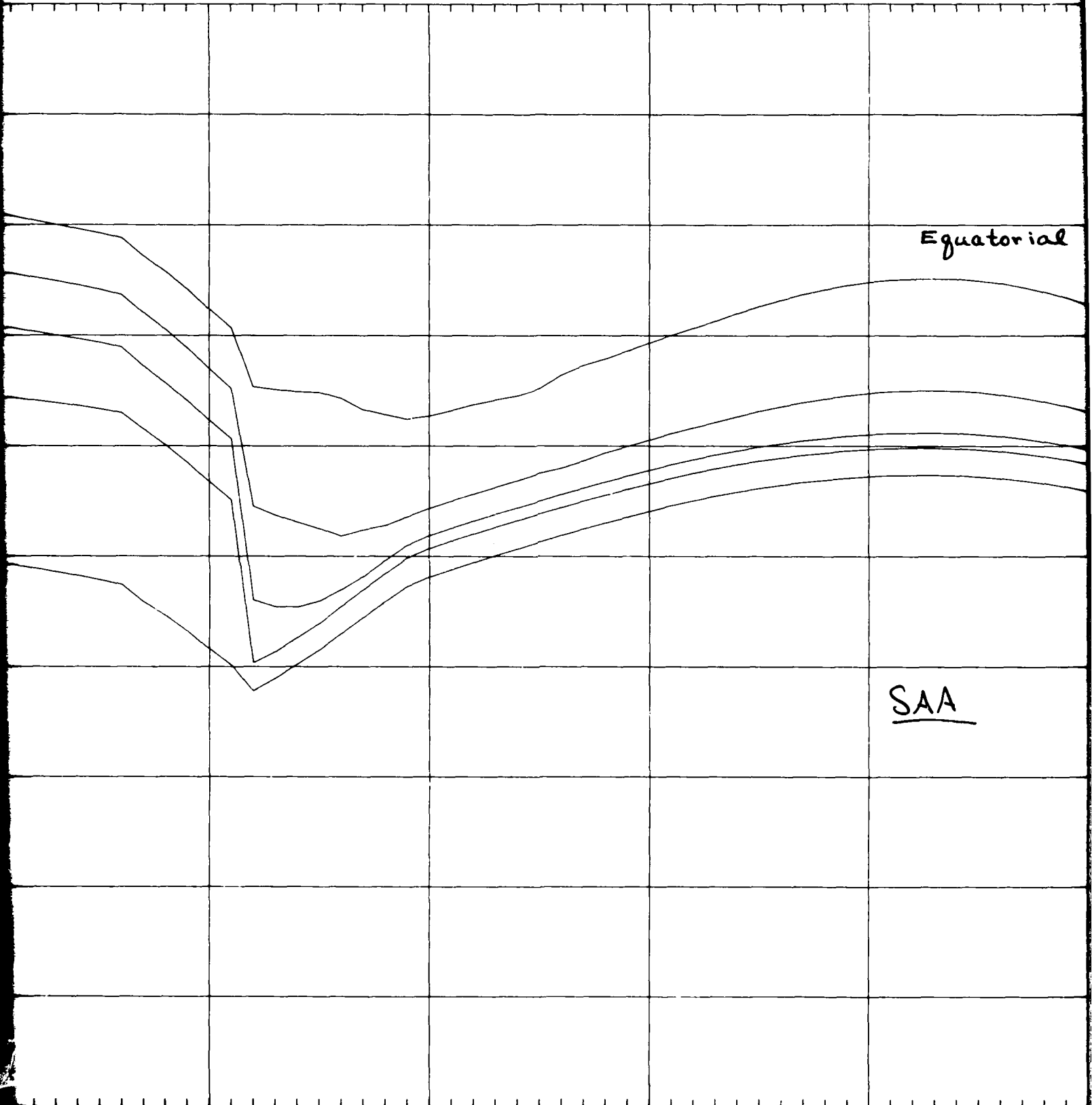
DOSE AT CENTER OF ALUMINUM SPHERES

Southern Horn



4

NUM SPHERES



Equatorial

SAA

0  
MINUTES)

120.0

130.0

140.0

150.0

160.0

ORBIT: NAV  
60 DGR/

EPOCH: 19

MODELS:  
FIELD: B  
TRAPPED  
INNER ZN  
OUTER ZN  
MISSION DU  
EVALUATION  
UN FACTORS

Northern  
Horn

Equatorial Region

SAA

STOP TIME

150.0 160.0 170.0 180.0 190.0

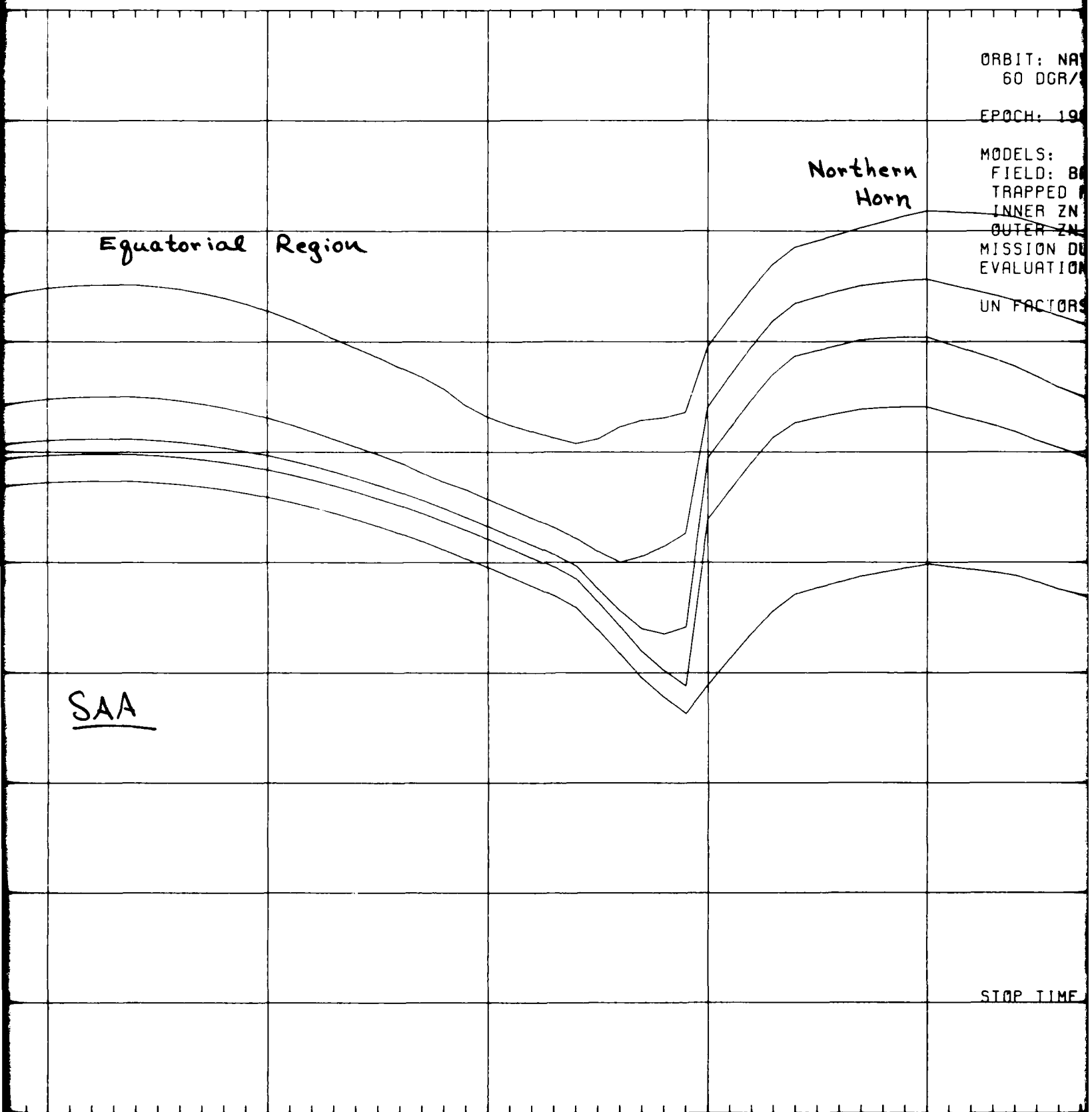


Figure 145

ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
~~OUTER ZN ELEC: AE17 L0~~  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern  
Horn

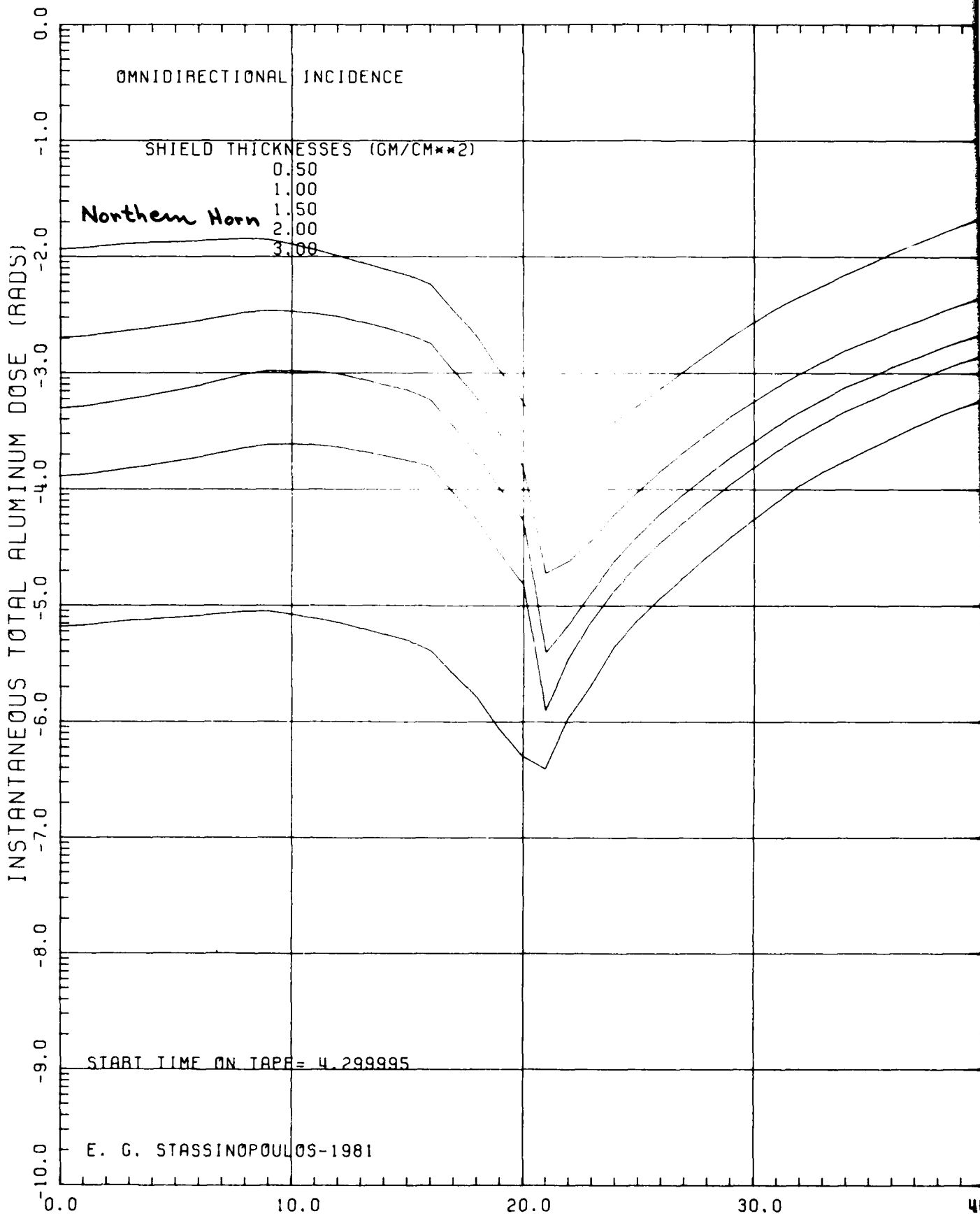
STOP TIME ON TAPE = 7.716662

NASA-GSFC

190.0

200.0

210.0





Equatorial Region

SAA

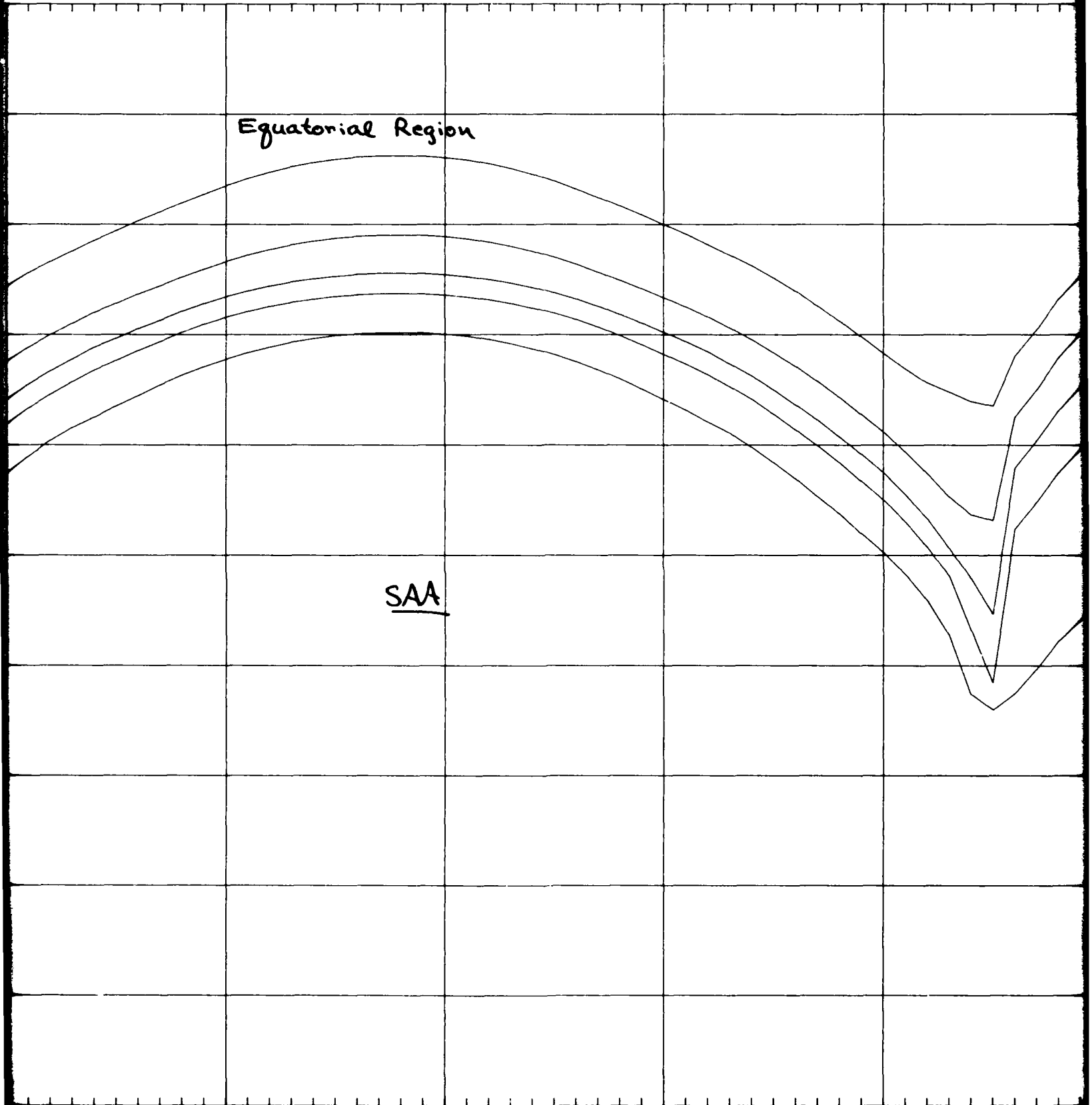
0

40.0

50.0

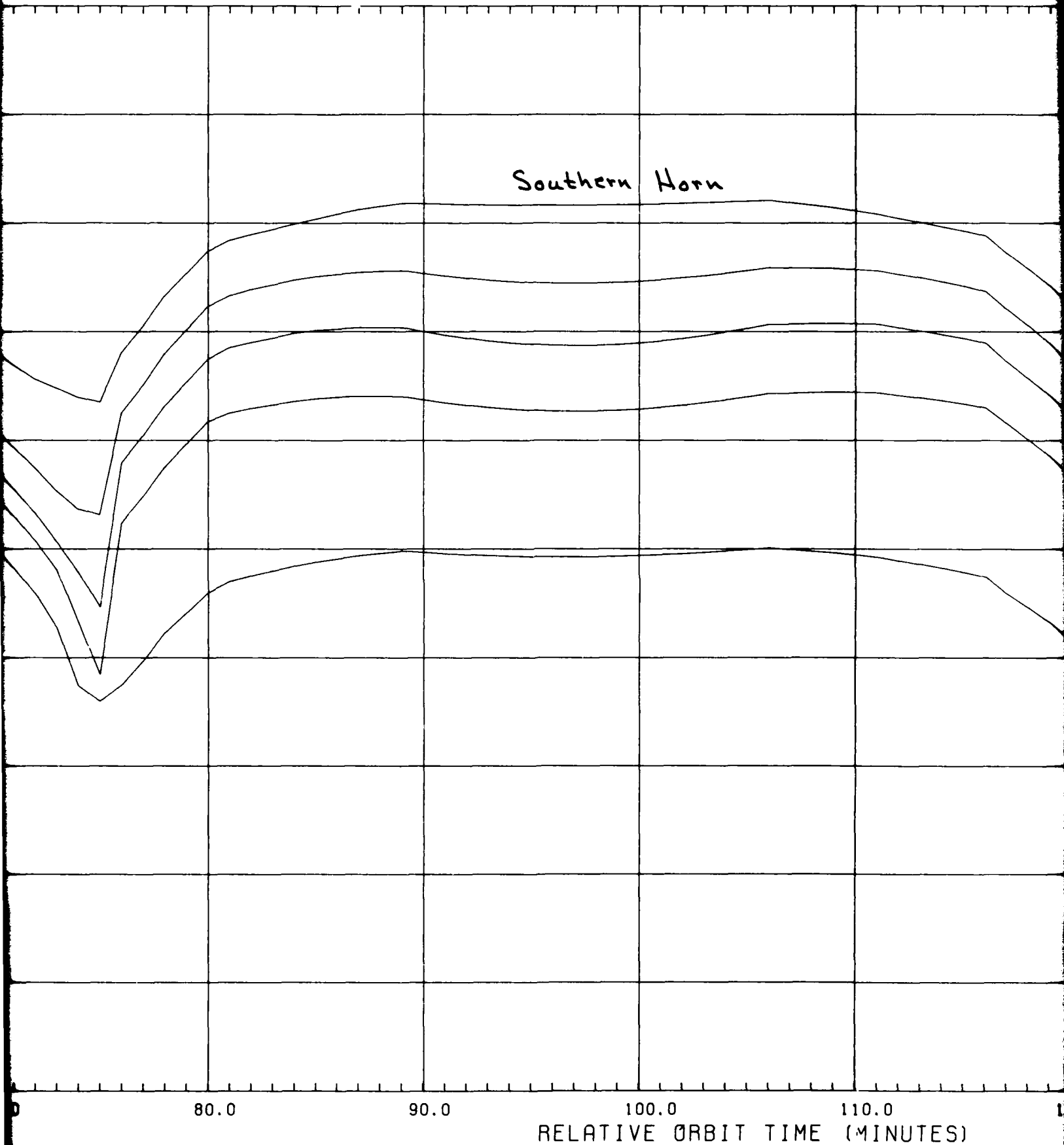
60.0

70.0



DOSE AT CENTER OF ALUMINUM SPHERES

Southern Horn



INUM SPHERES

14

Equatorial R

SAA

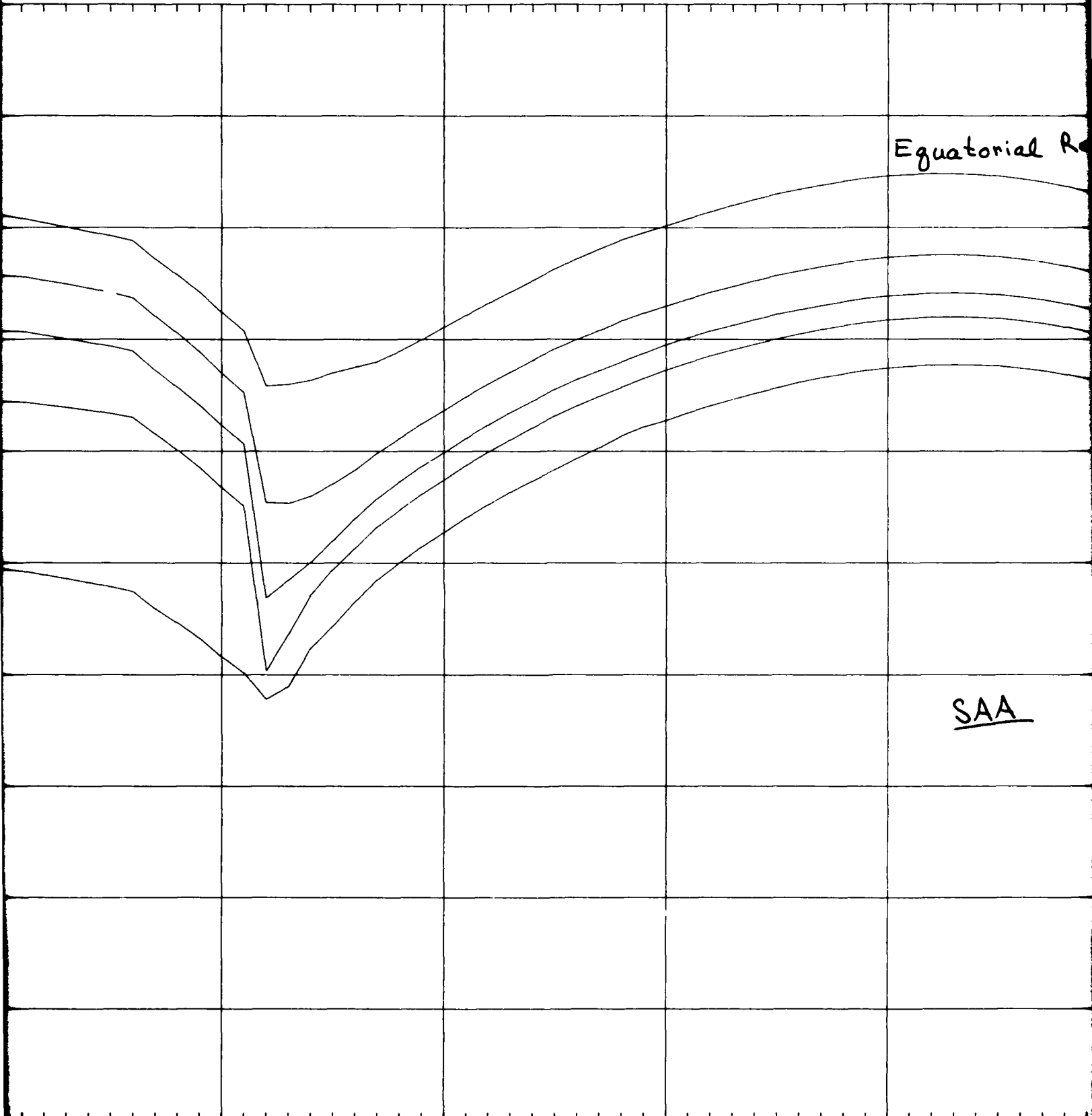
0.0  
(MINUTES)

120.0

130.0

140.0

150.0



ORBIT: NAVEL  
60 DGR/518

EPOCH: 1989

MODELS:  
FIELD: BARR  
TRAPPED PRO  
INNER ZN EL  
OUTER ZN EL  
MISSION DUR  
EVALUATION P

UN FACTORS:

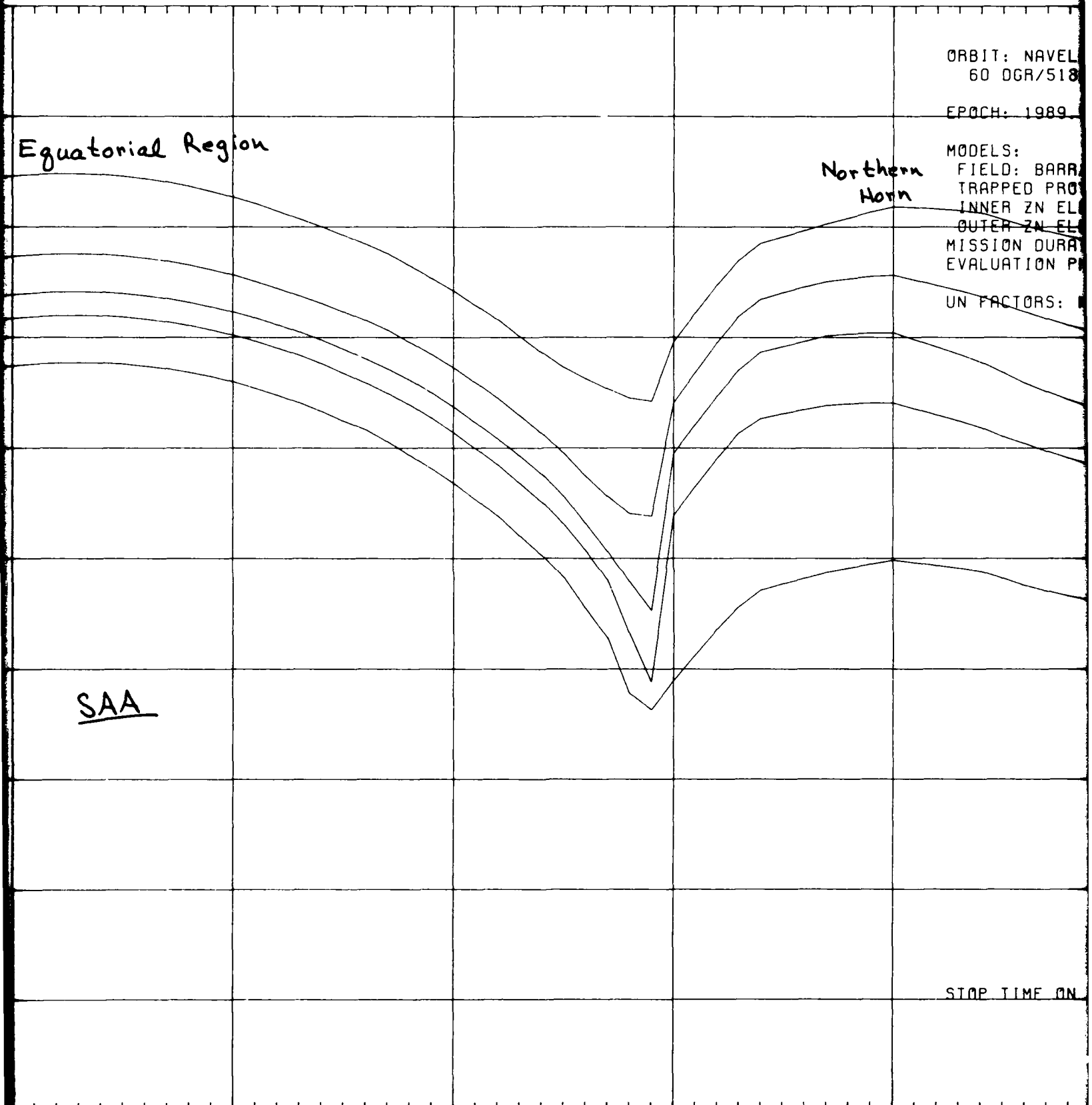
Equatorial Region

Northern  
Horn

SAA

STOP TIME ON

0.0 160.0 170.0 180.0 190.0



6

Figure 146

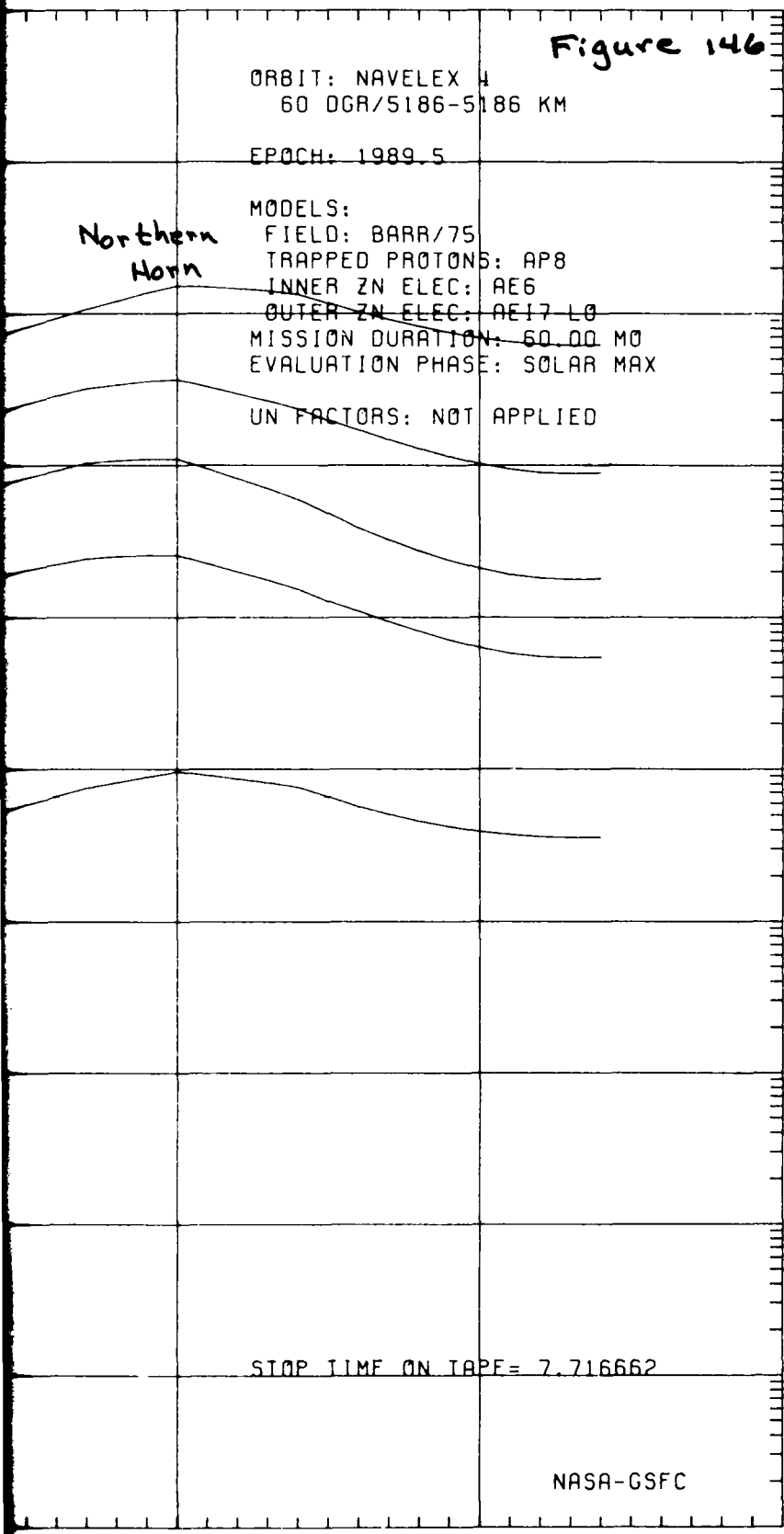
ORBIT: NAVELEX 4  
60 DGR/5186-5186 KM

EPOCH: 1989.5

Northern  
Horn

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
~~OUTER ZN ELEC: AE17 L0~~  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

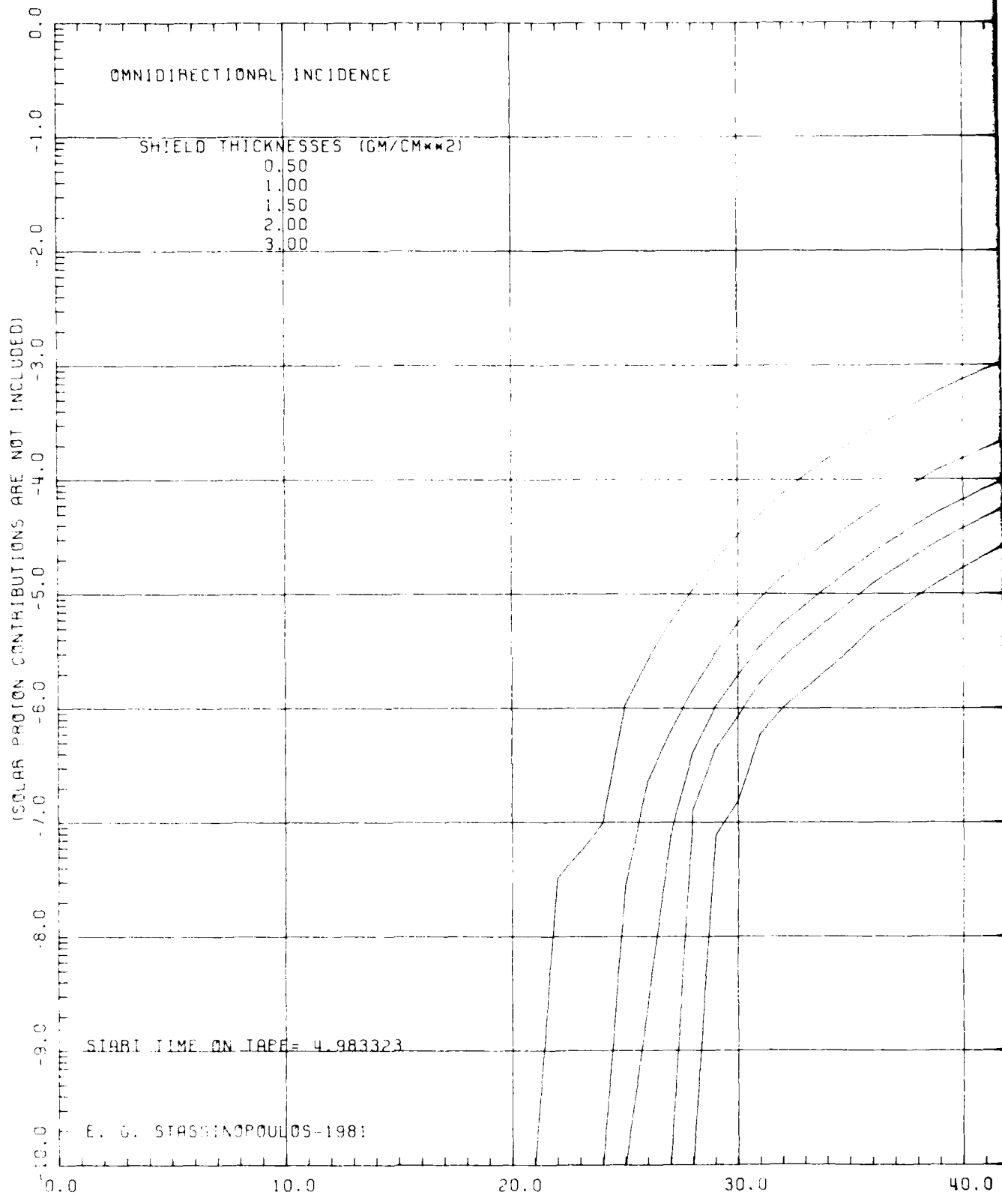


STOP TIME ON TAPE = 7.716662

NASA-GSFC

190.0                      200.0                      210.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)



Equatorial Region

SAA

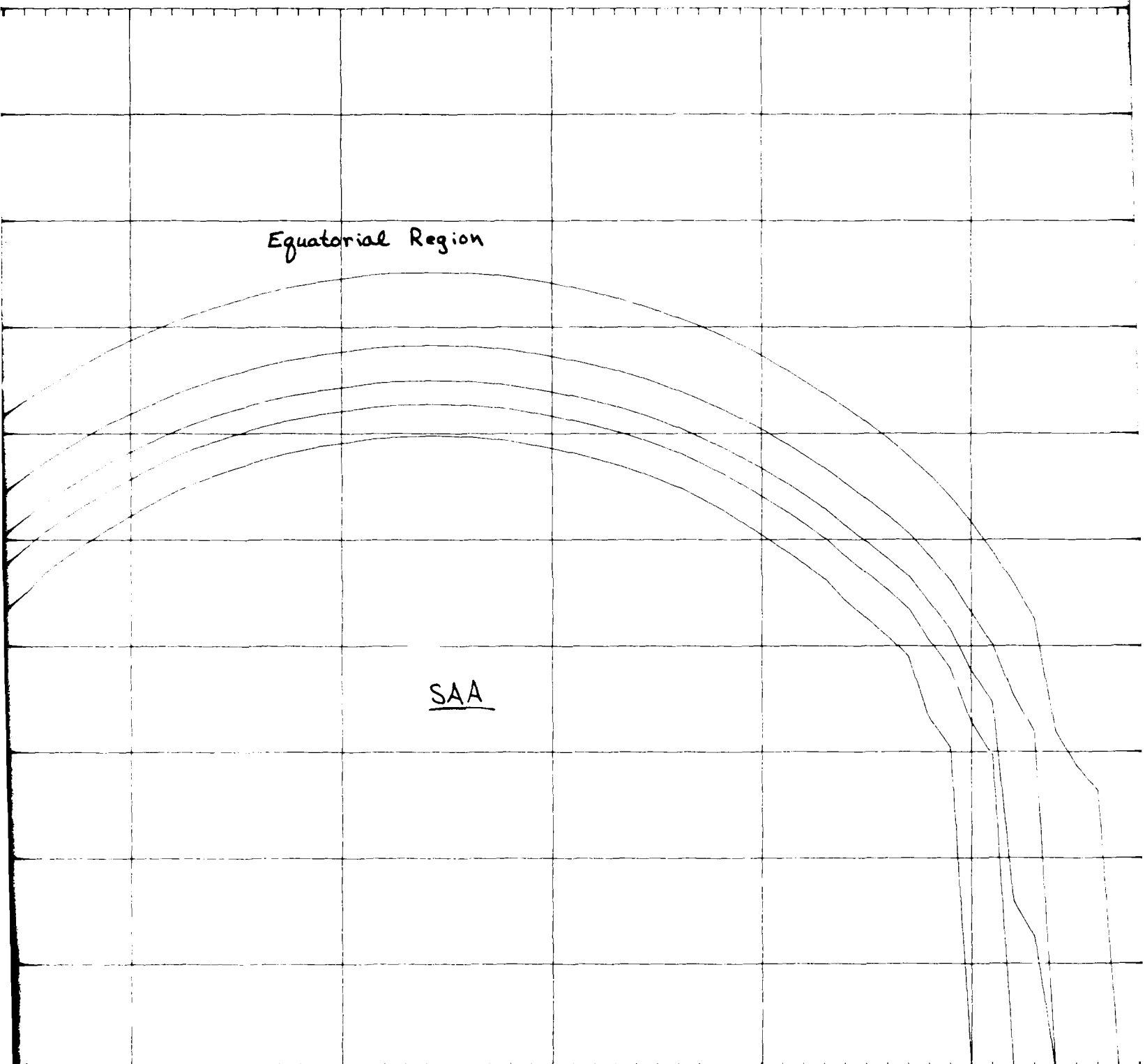
40.0

50.0

60.0

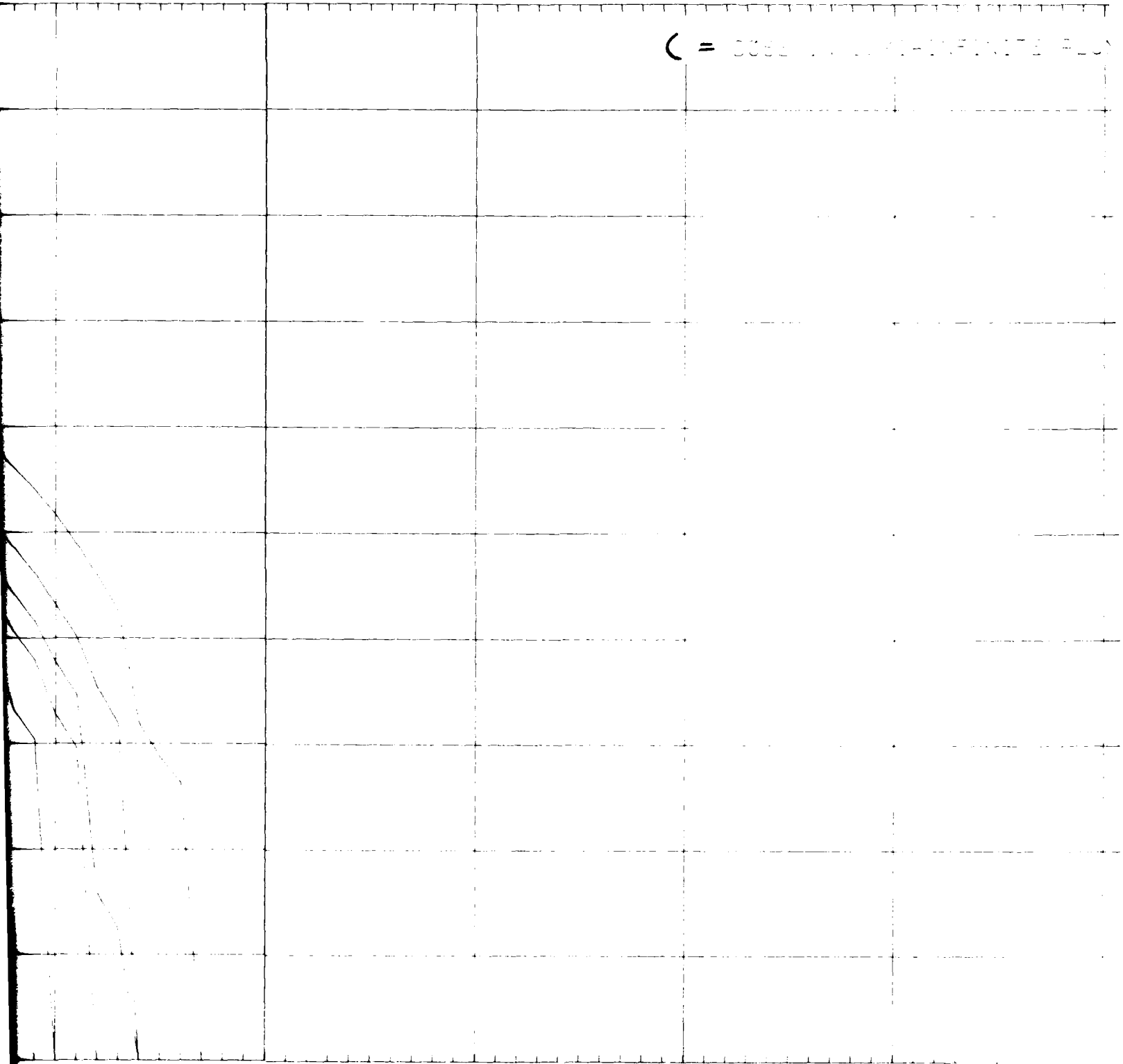
70.0

80.0



DOSE AT TRANSMISSION SURFACE OF FINITE AL

$C = \text{DOSE AT TRANSMISSION SURFACE OF FINITE AL}$



80.0 90.0 100.0 110.0 120.0

RELATIVE DOSE



AD-A141 849

ORBITAL RADIATION STUDY FOR INCLINED CIRCULAR  
TRAJECTORIES(U) NATIONAL AERONAUTICS AND SPACE  
ADMINISTRATION GREENBELT MD GO.. E G STASSINOPOULOS  
NOV 81 NASA-GSFC-X-601-81-28

5/5

UNCLASSIFIED

F/G 22/3

NL

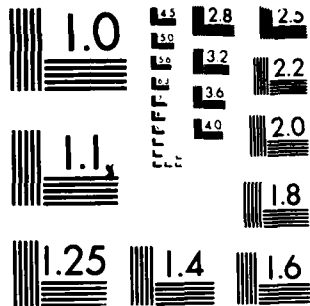
END

DATE

FORMED

7-84

DTHC



MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS 1963-A

4

FINITE ALUMINUM SLAB SHIELDS

FINITE ALUMINUM MEDIUM )

Equator

SAA

130.0

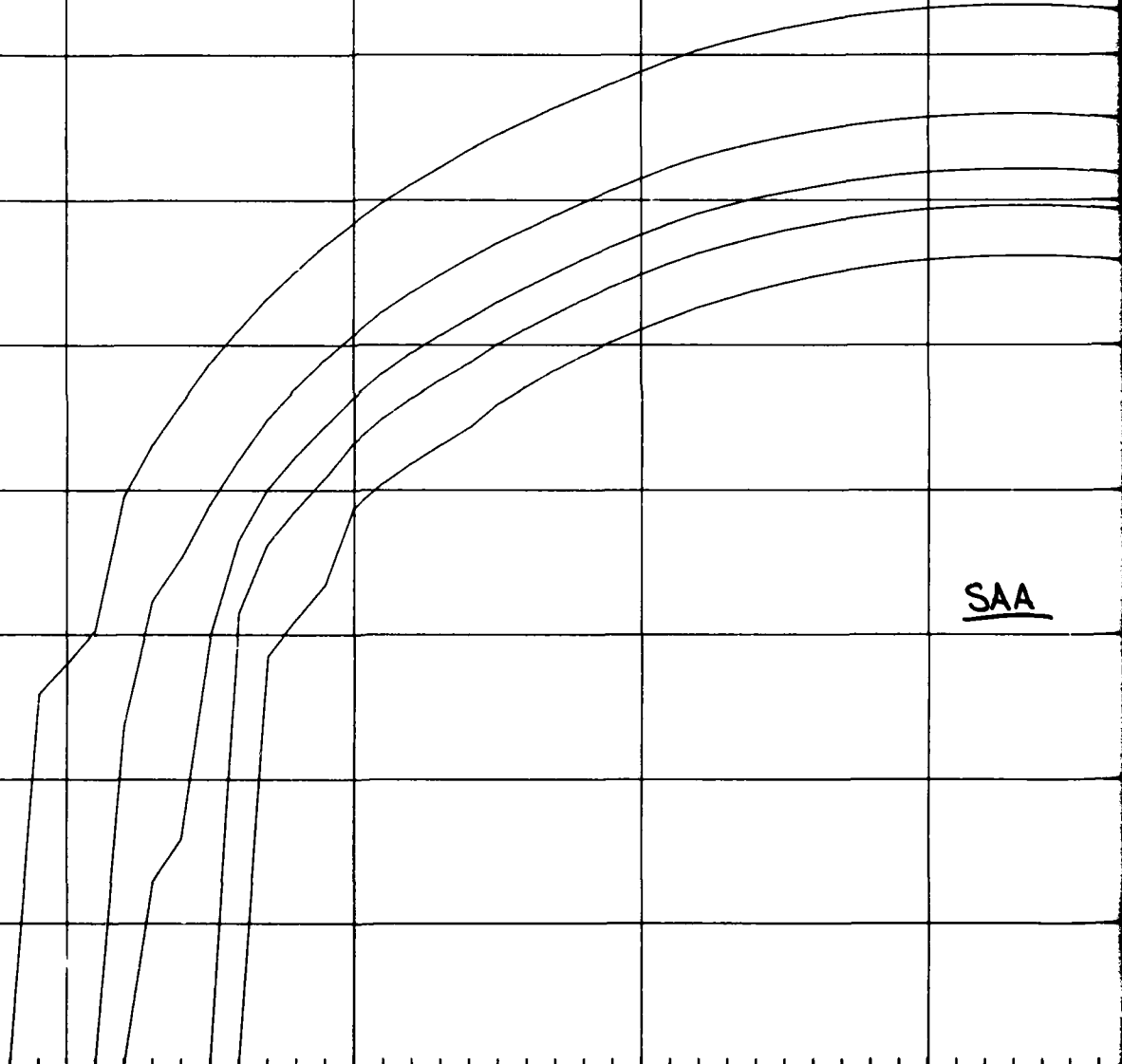
140.0

150.0

160.0

170.0

(MINUTES)



1  
5

Equatorial Region

SAA

170.0

180.0

190.0

200.0

210.0

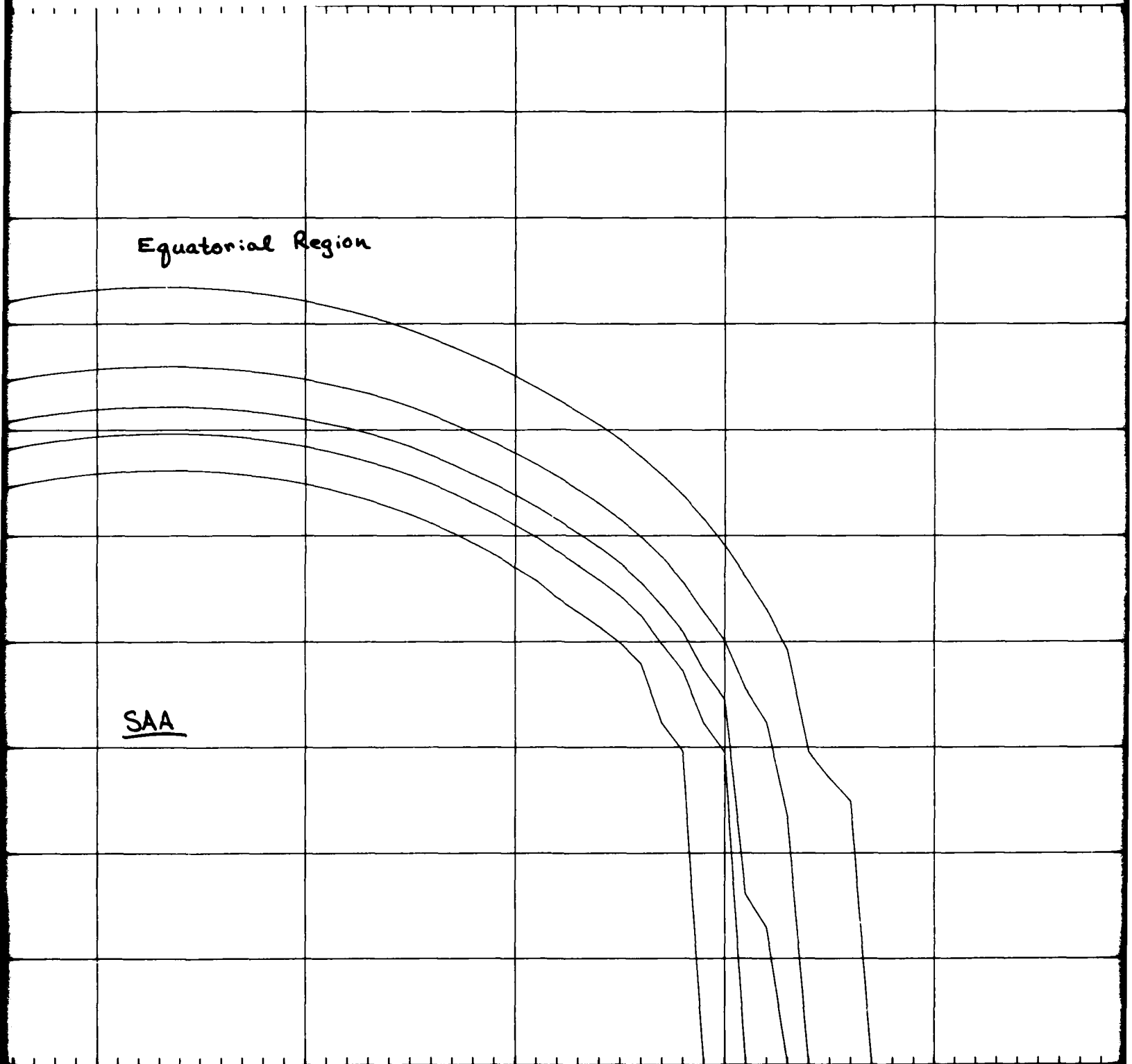


Figure 147

ORBIT: NAVELEX 5  
60 DGR/6389-6889 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

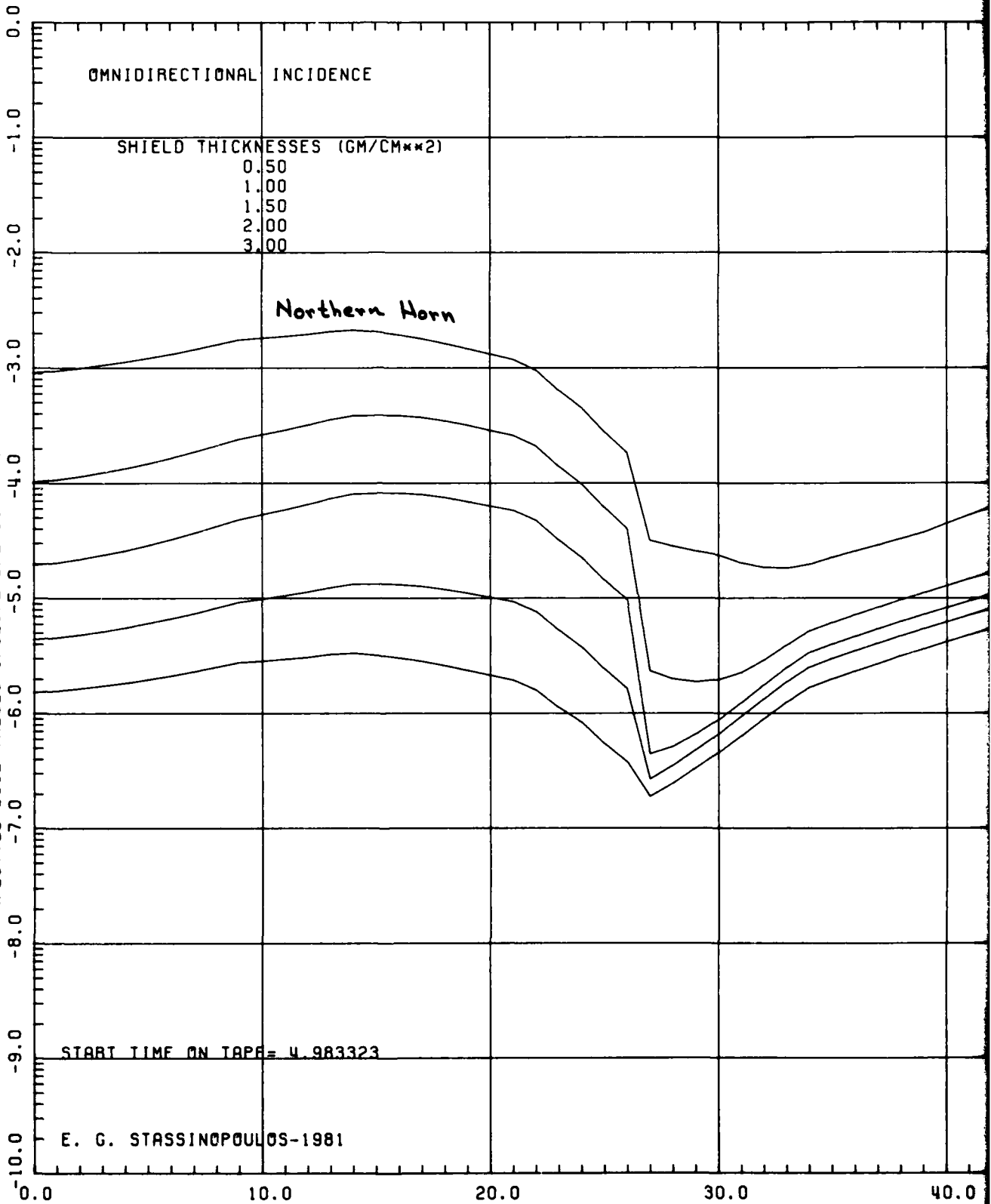
STOP TIME ON TAPE= 8.933318

NASA-GSFC

210.0                      220.0                      230.0                      240.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



'2

Equatorial Region

SAA

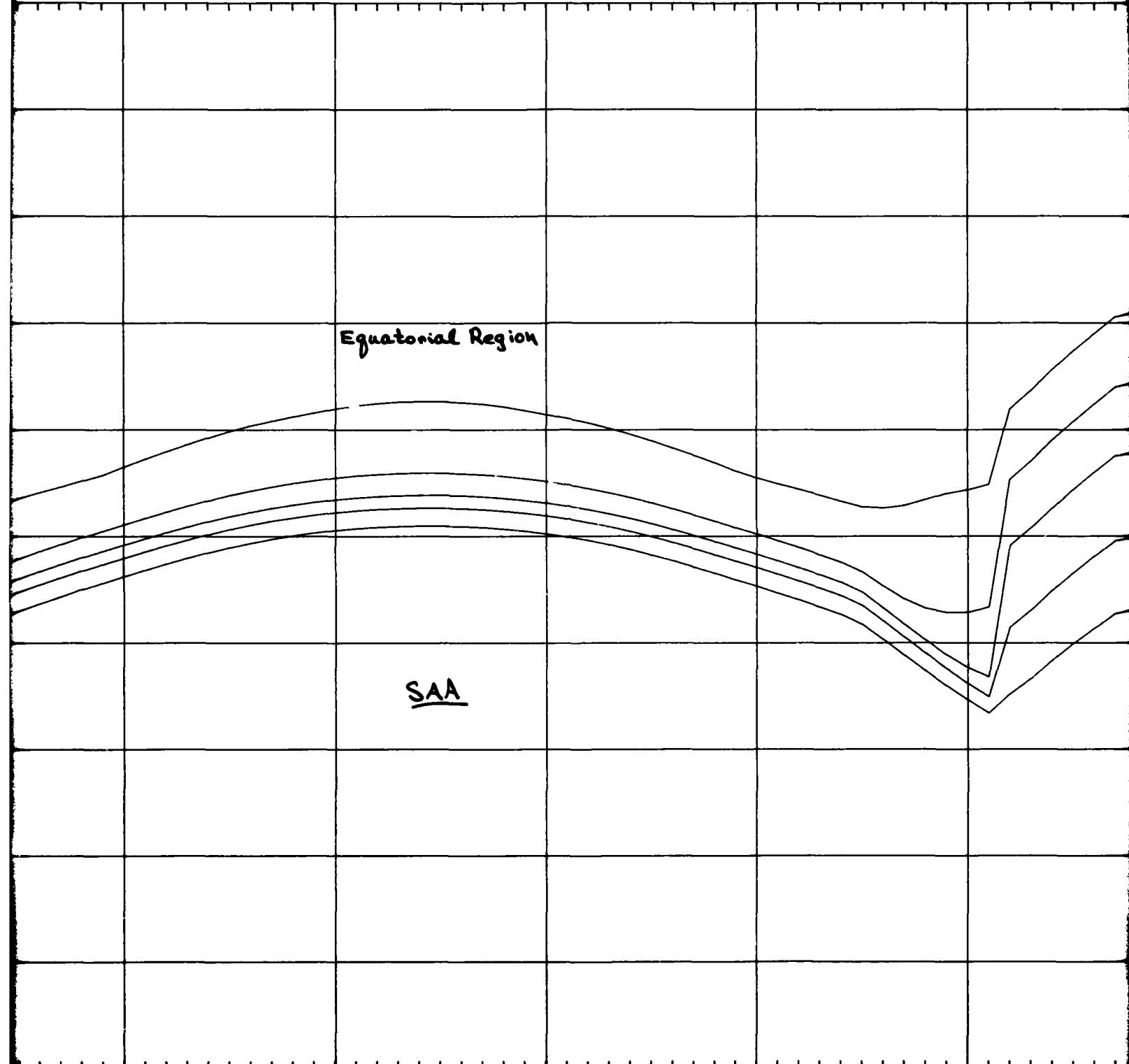
40.0

50.0

60.0

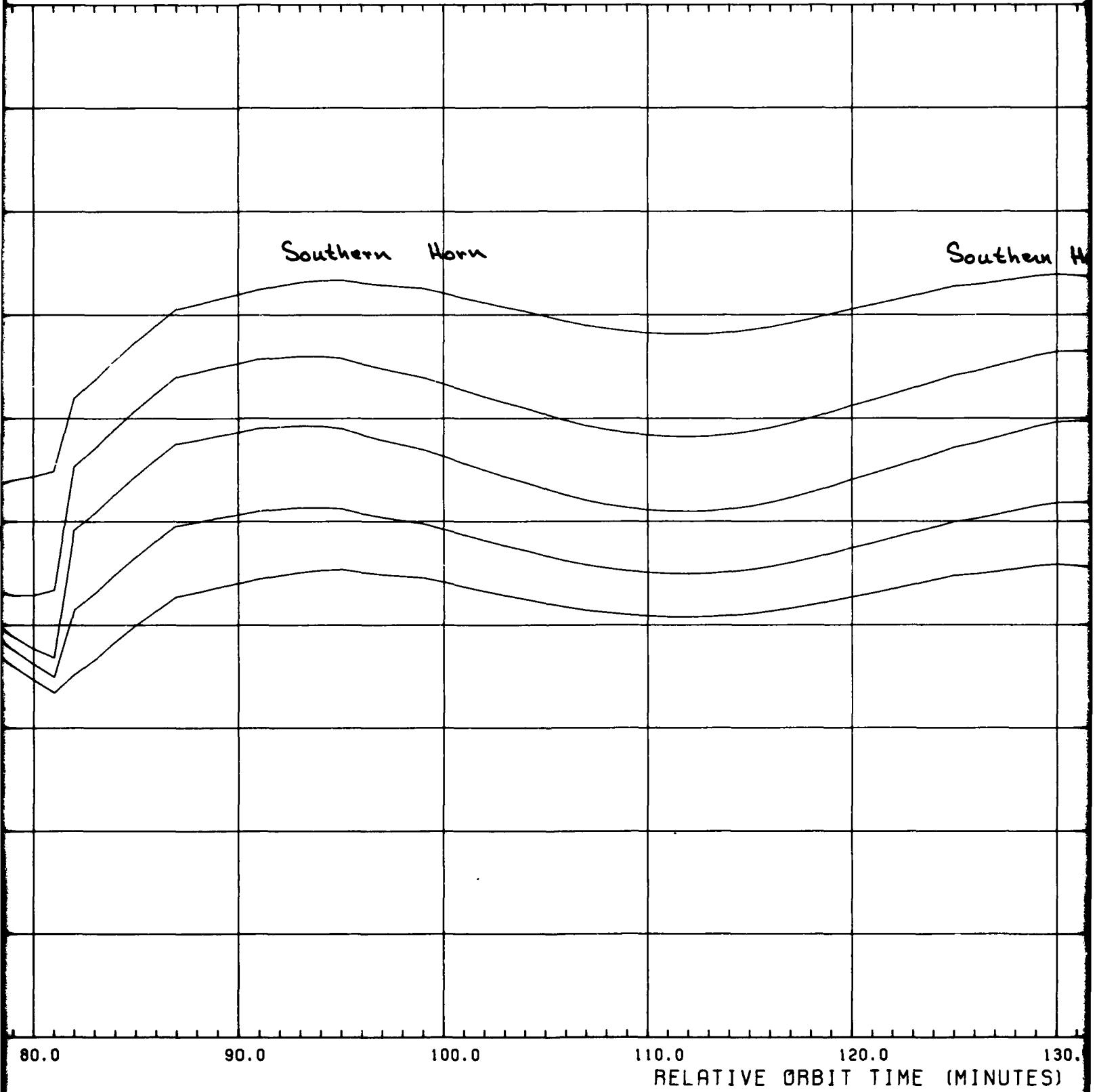
70.0

80.0



3

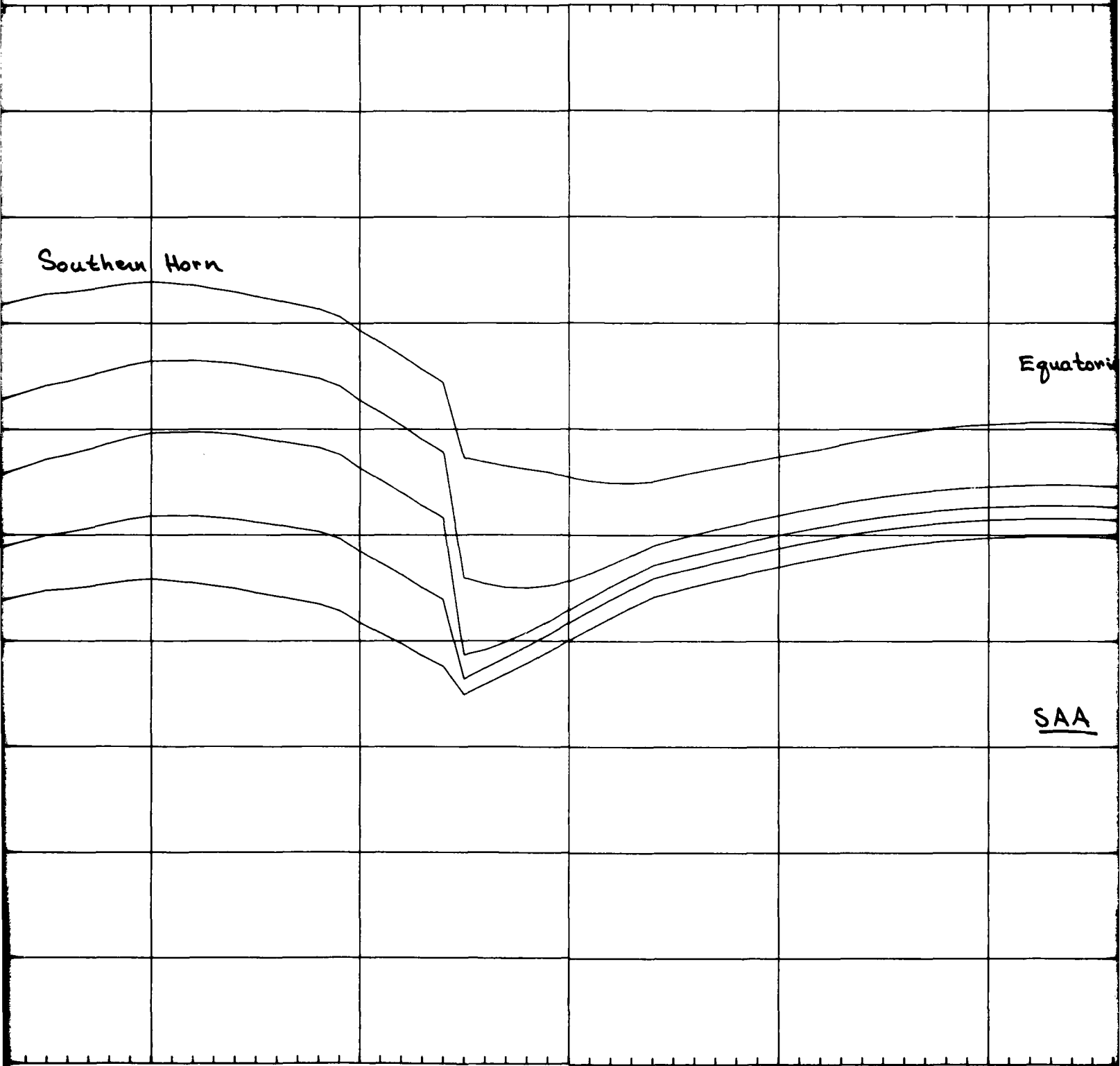
DOSE AT TRANSMISSION SURFACE OF FINITE ALU





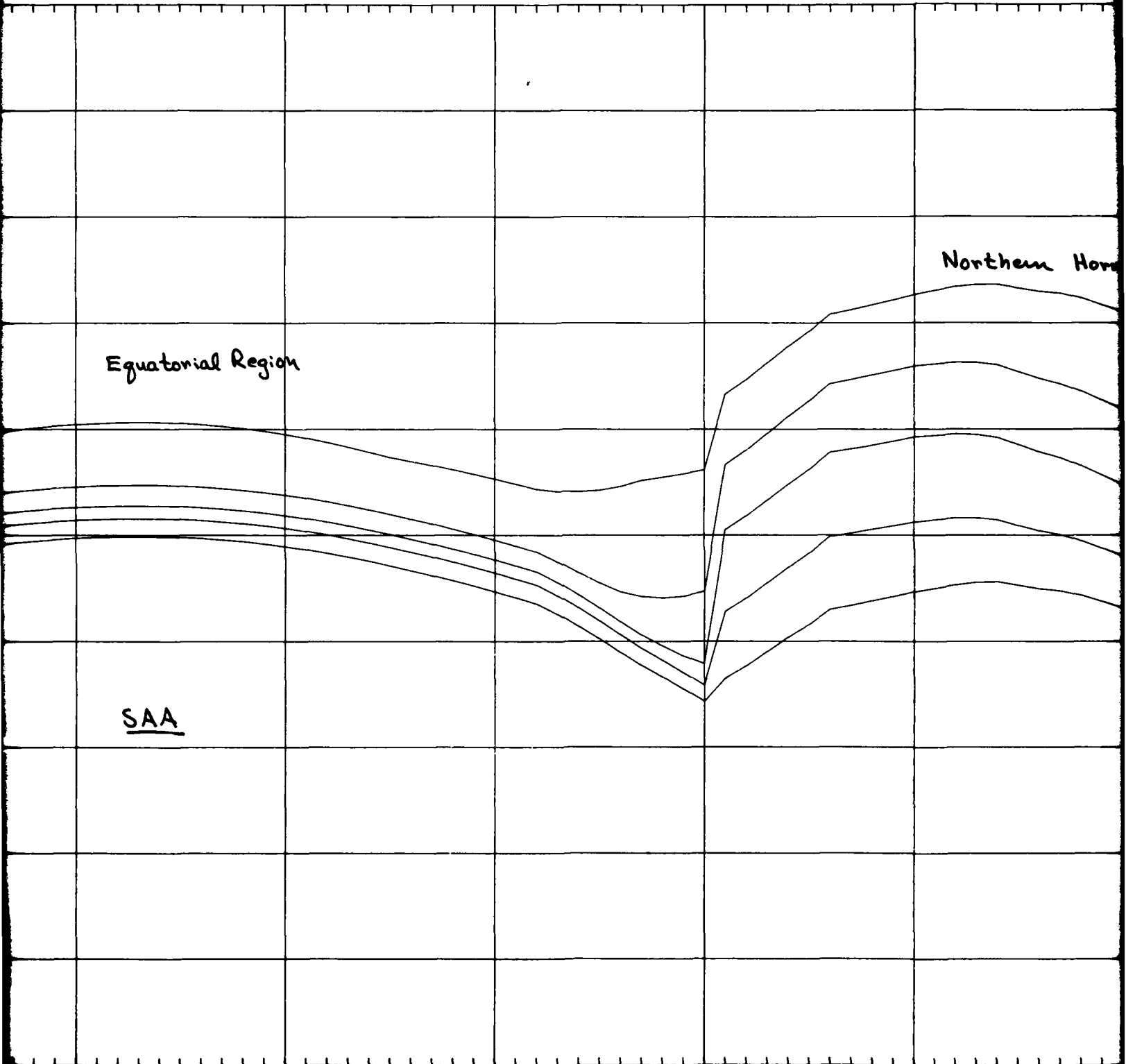
'4

FINITE ALUMINUM SLAB SHIELDS



130.0 140.0 150.0 160.0 170.0  
(MINUTES)

'5



170.0

180.0

190.0

200.0

210.0

2

D-3

Figure 148

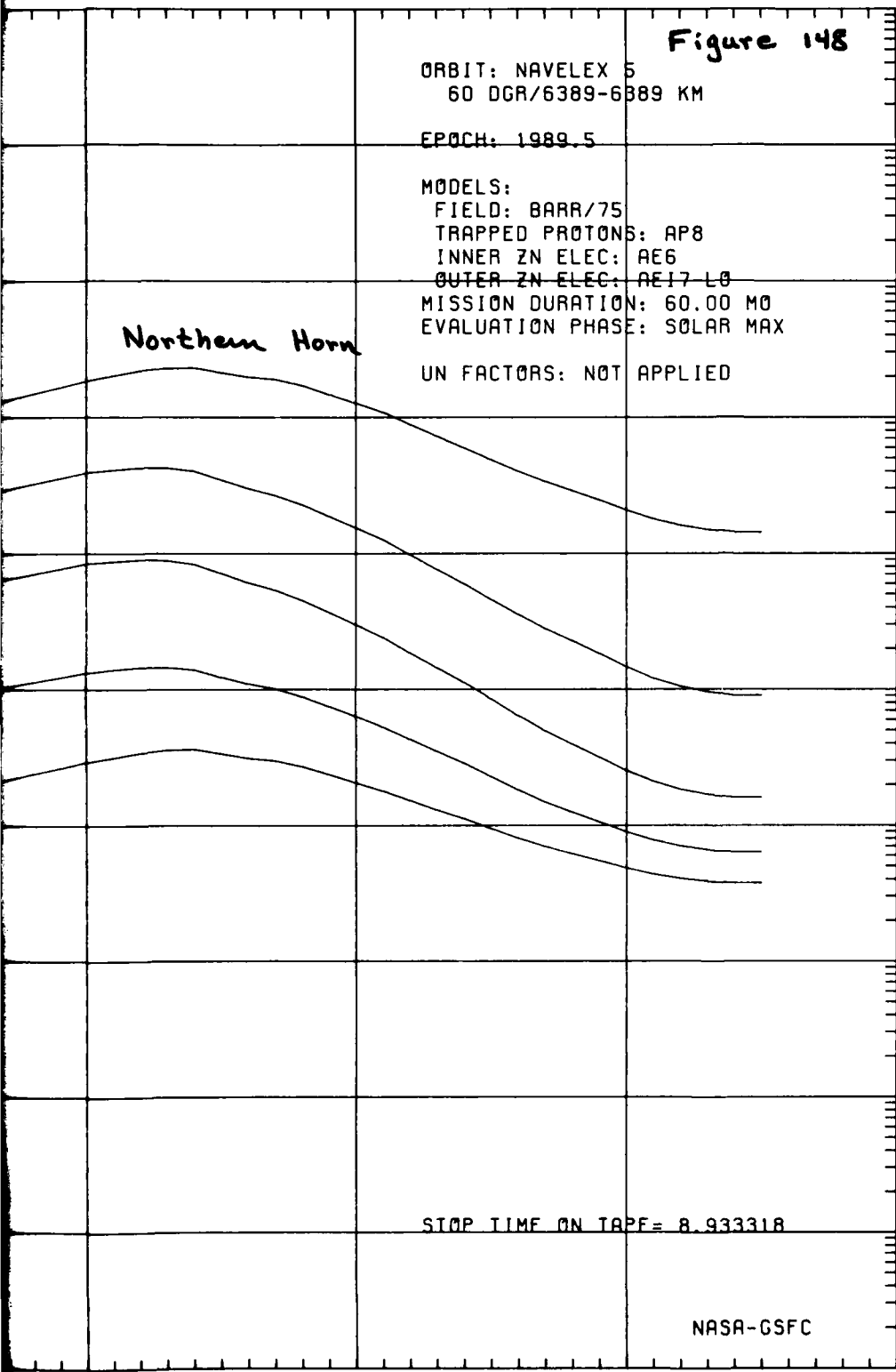
ORBIT: NAVELEX 5  
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

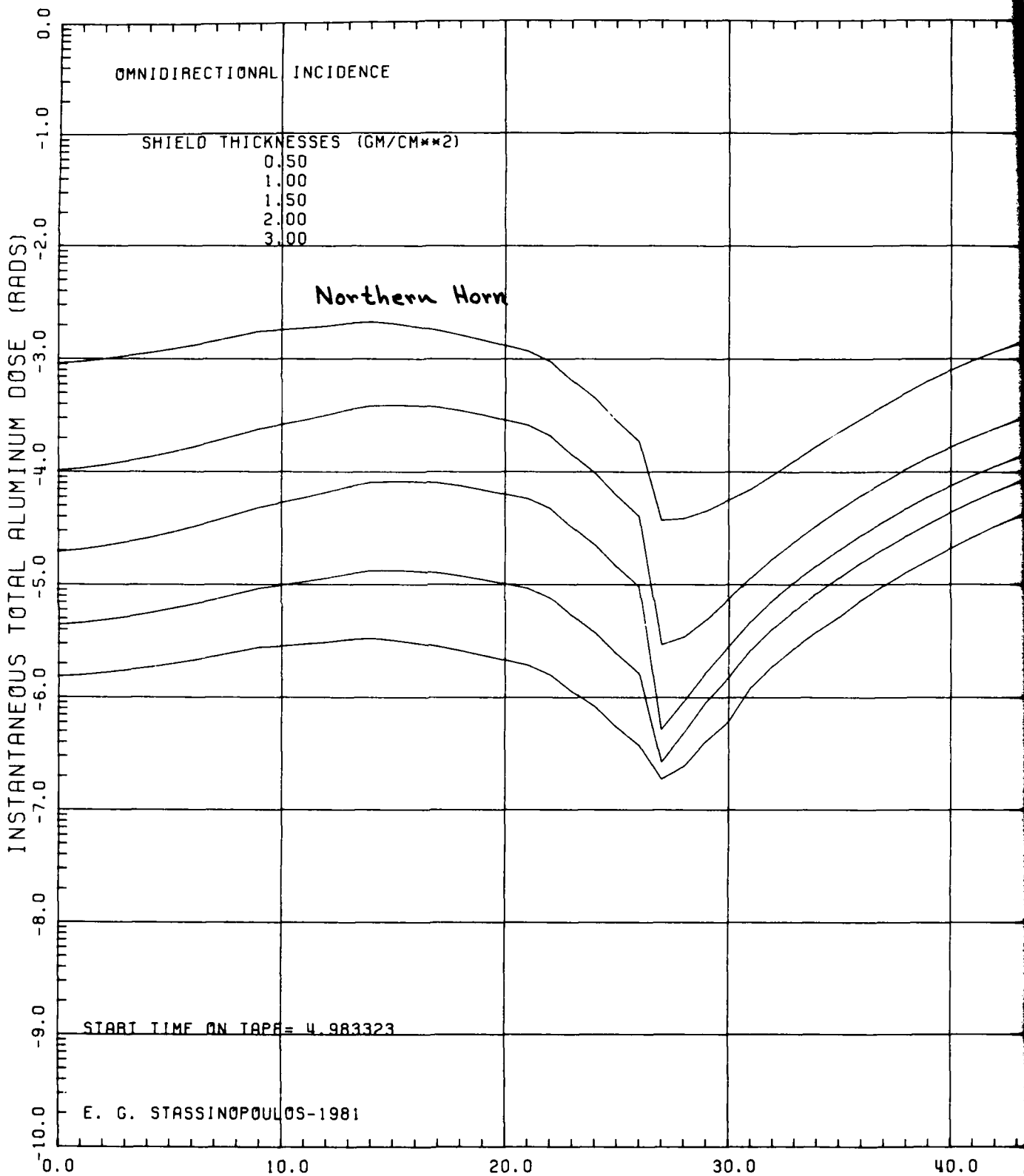
Northern Horn



STOP TIME ON TAPE = 8.933318

NASA-GSFC

210.0                      220.0                      230.0                      240.0



2

Equatorial Region

SAA

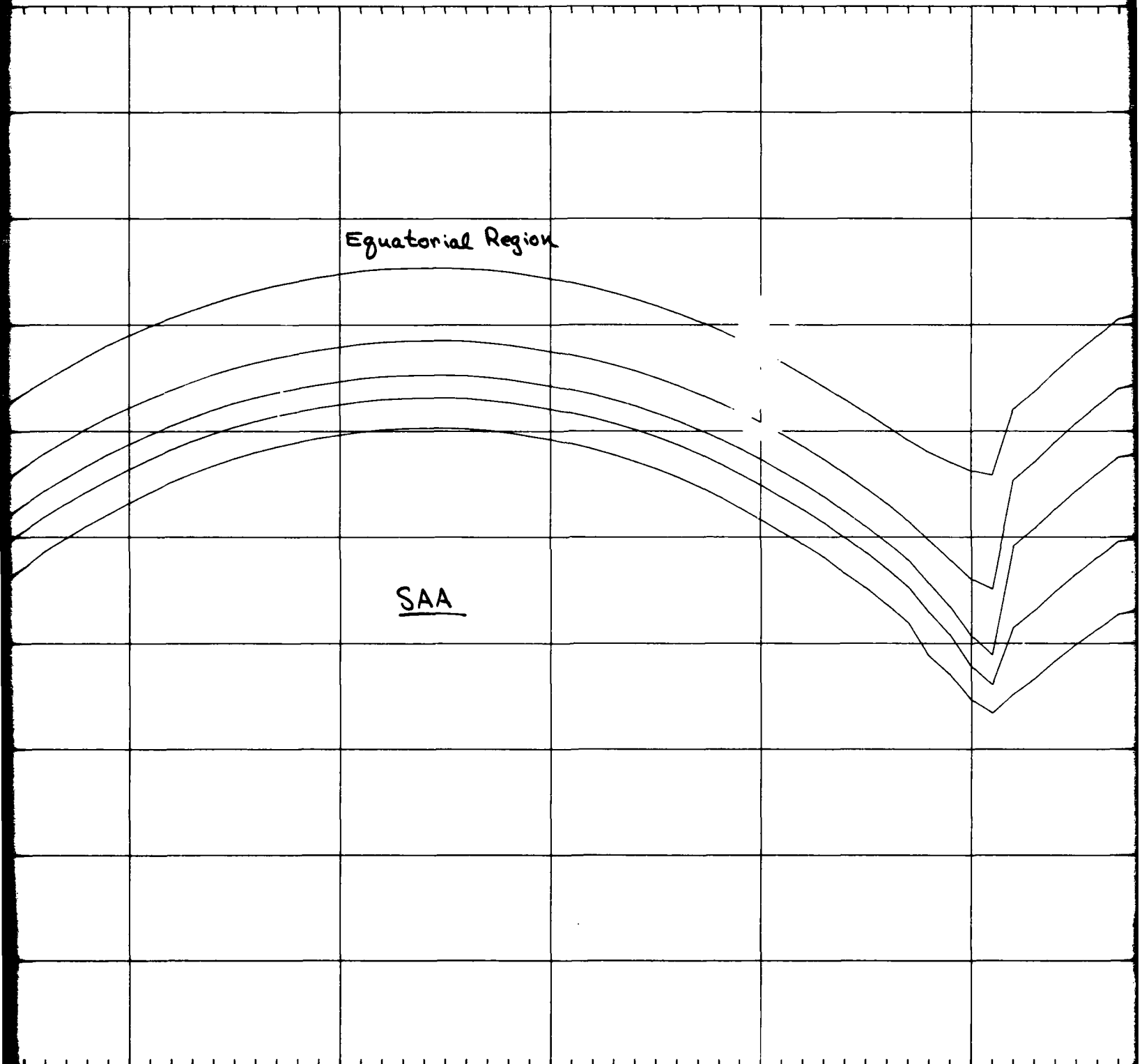
40.0

50.0

60.0

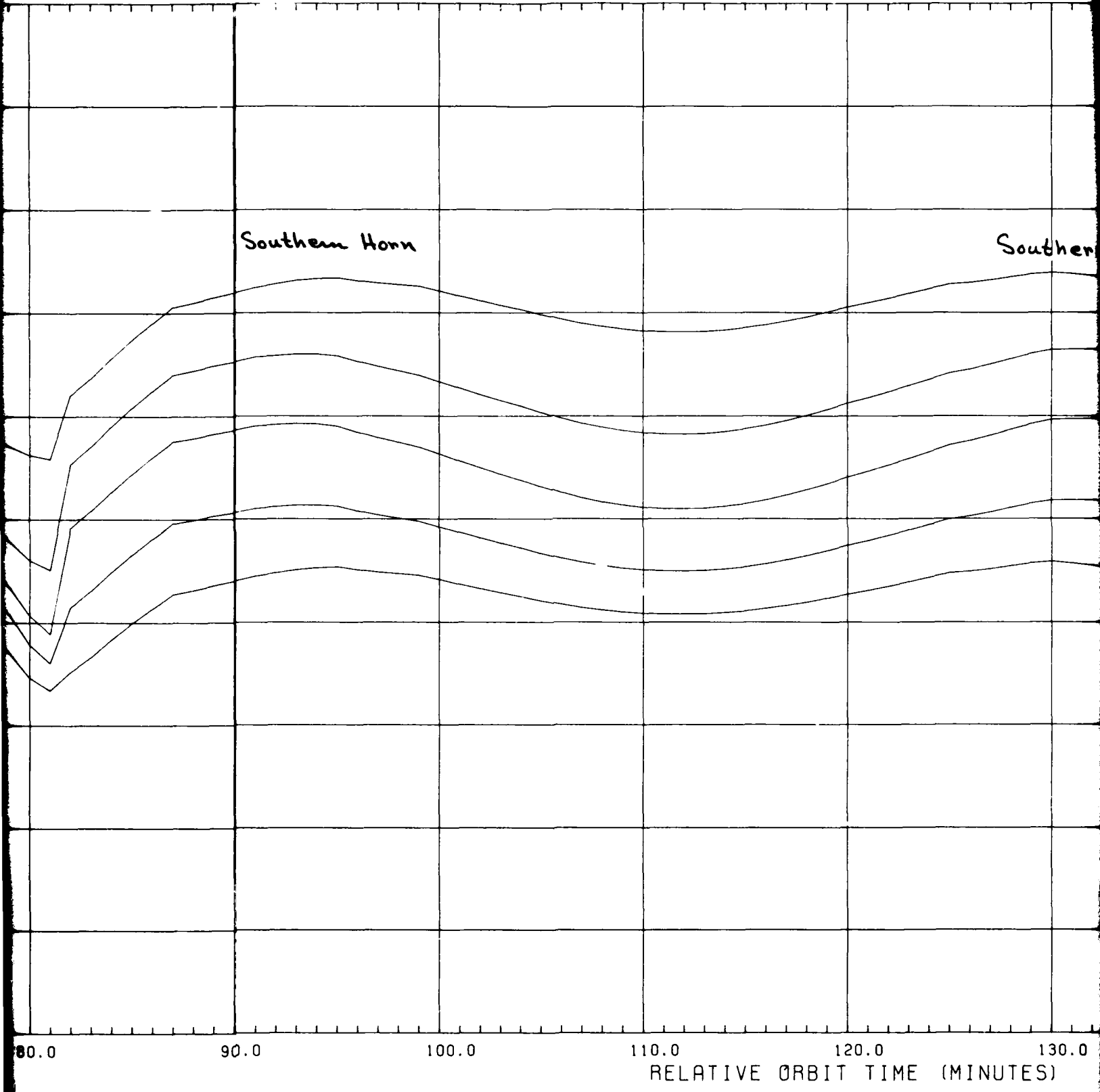
70.0

80.0



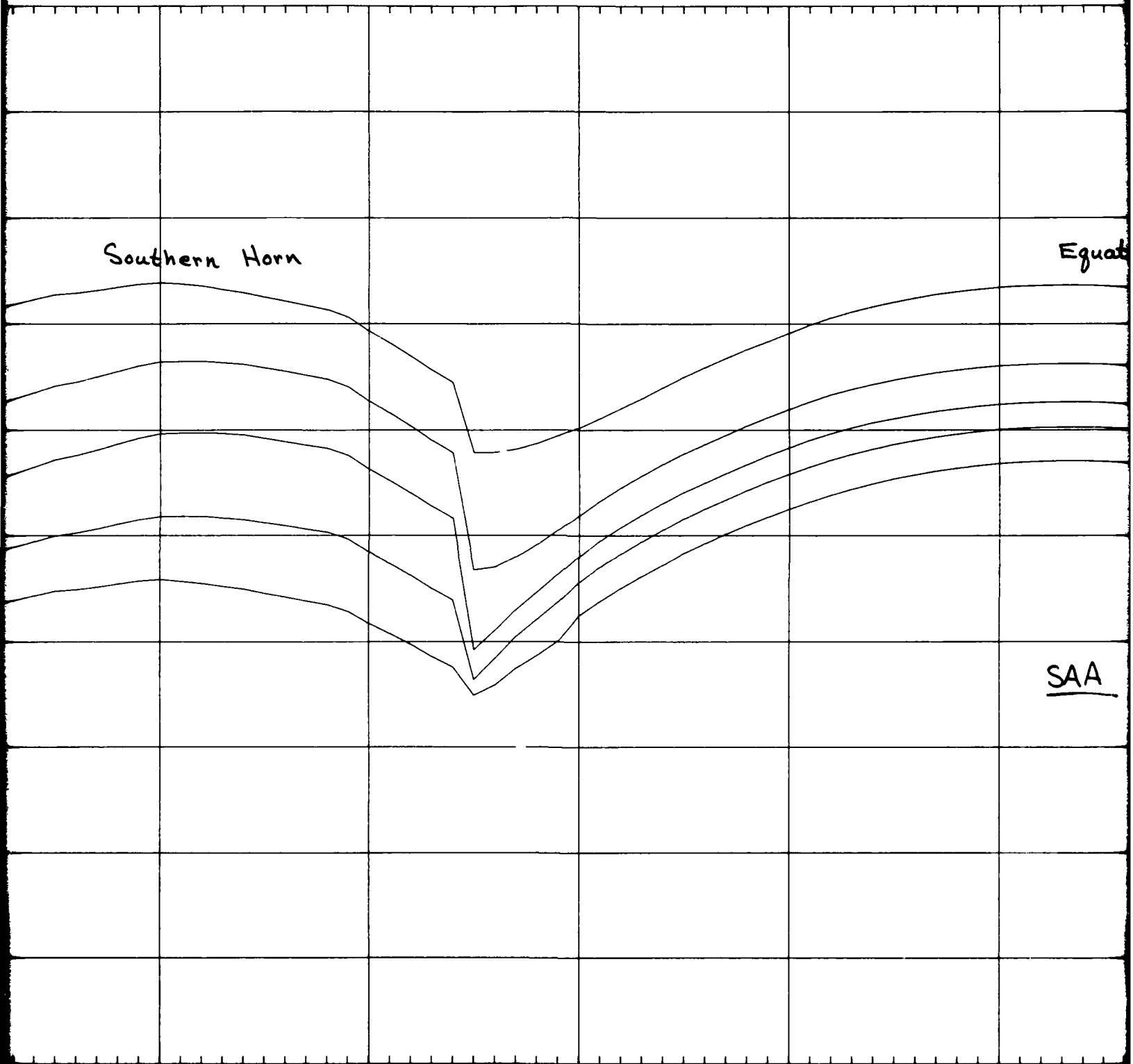
3

DOSE AT TRANSMISSION SURFACE OF FINITE ALUM



4

F FINITE ALUMINUM SLAB SHIELDS



E (MINUTES) 130.0 140.0 150.0 160.0 170.0

5

Equatorial Region

Northern Horn

SAA

170.0

180.0

190.0

200.0

210.0

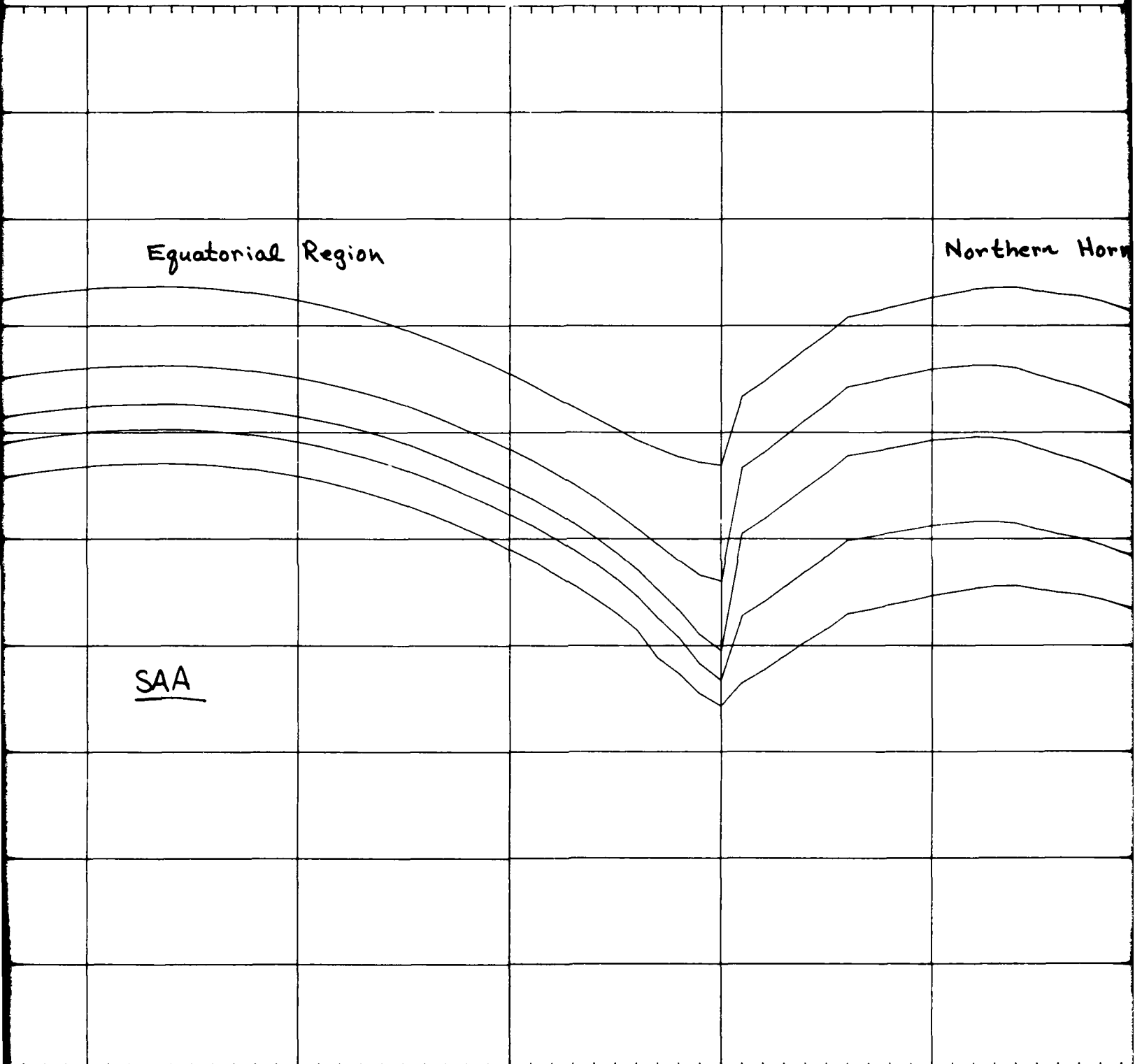




Figure 149

ORBIT: NAVELEX 5  
60 DGR/6389-6389 KM

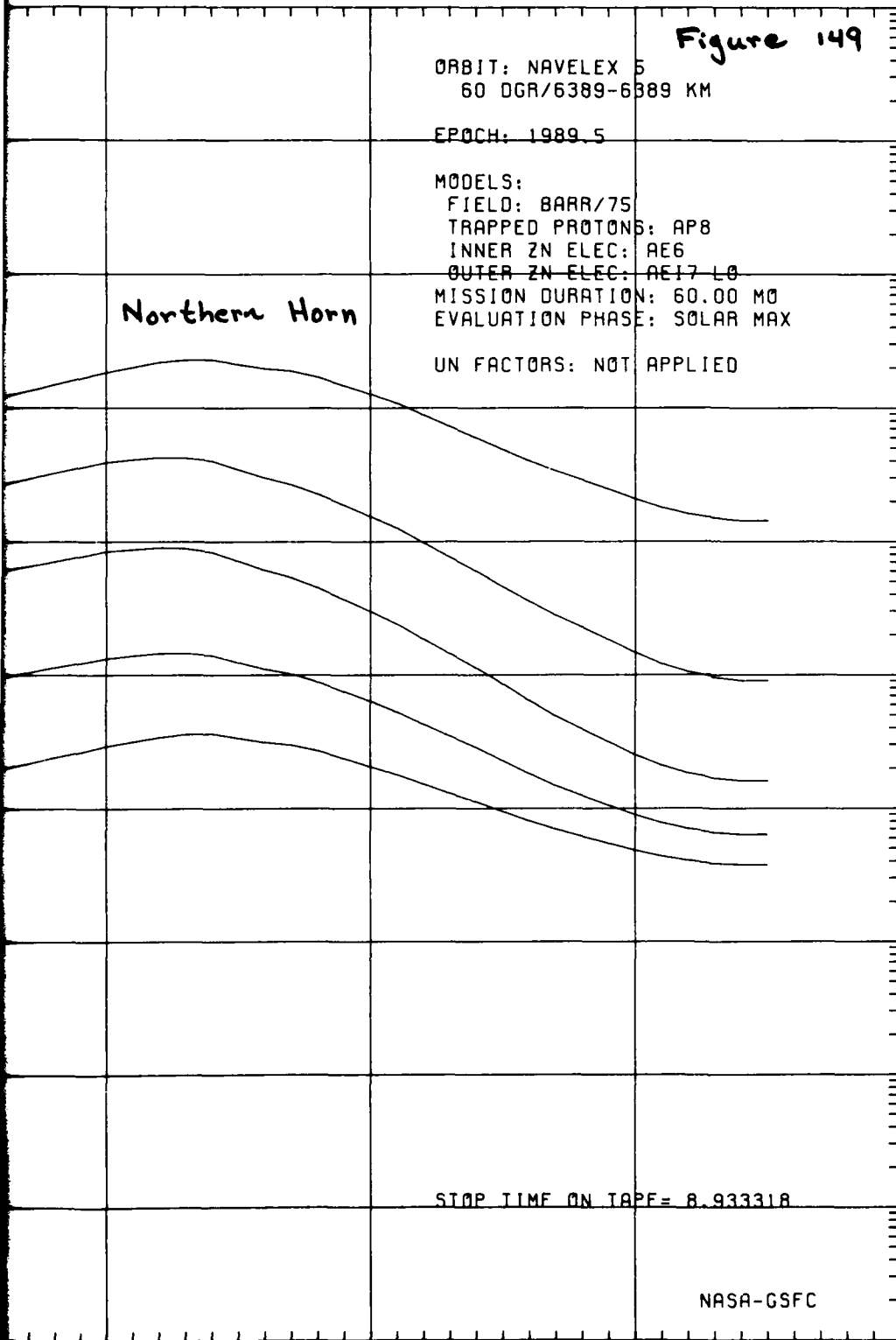
EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn



STOP TIME ON TAP = 8.933318

NASA-GSFC

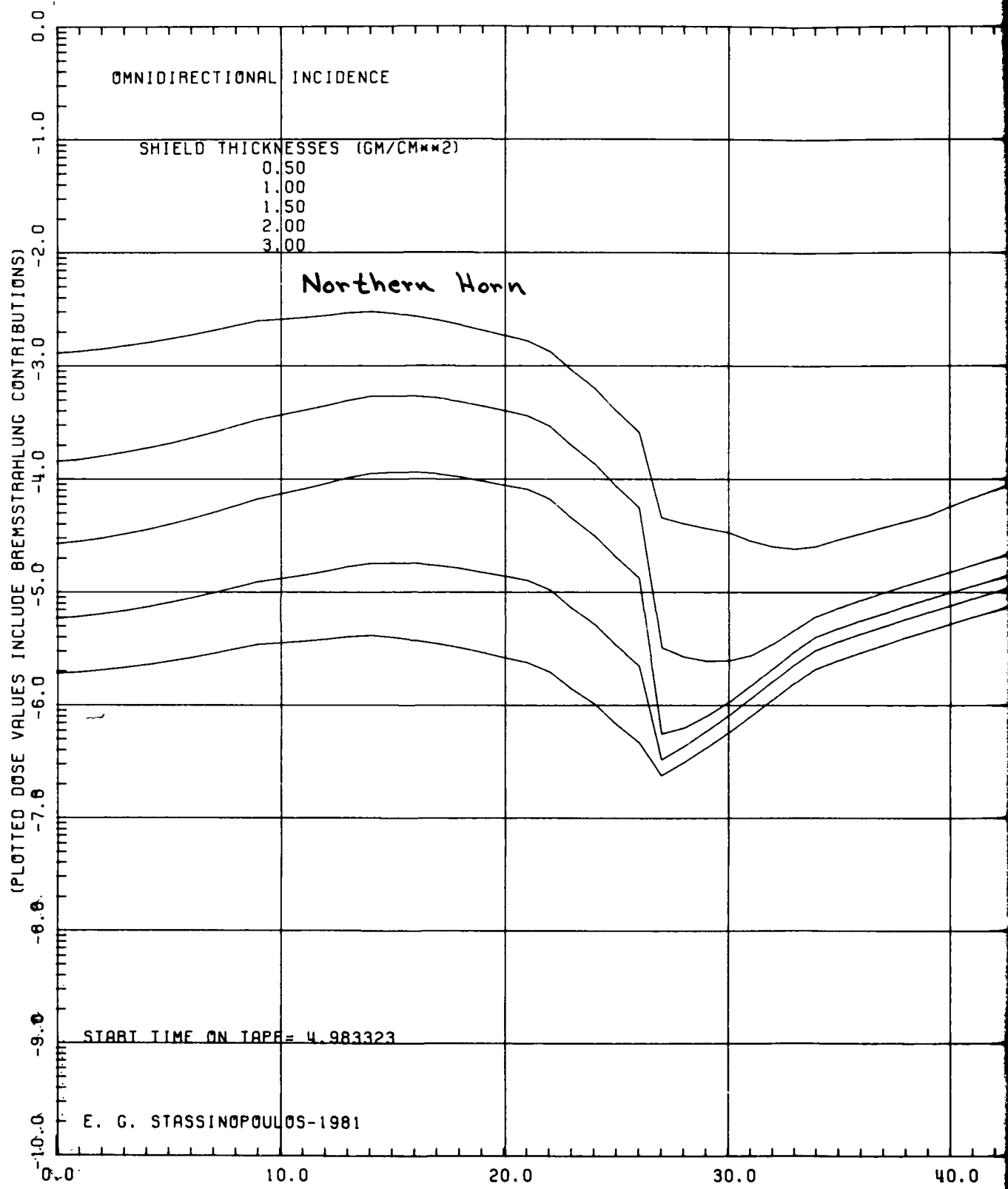
210.0

220.0

230.0

240.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)



2

Equatorial Region

SAA

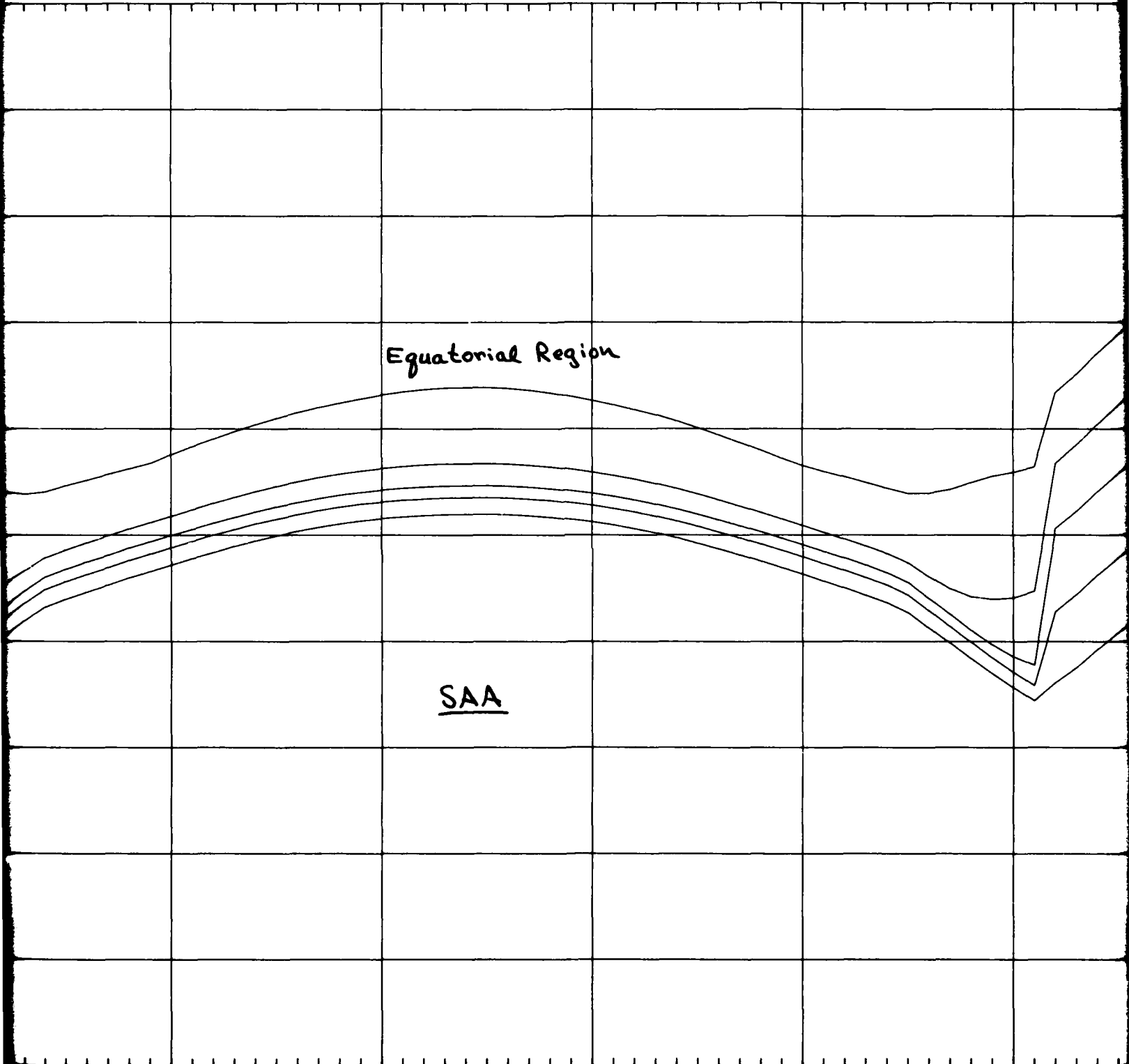
40.0

50.0

60.0

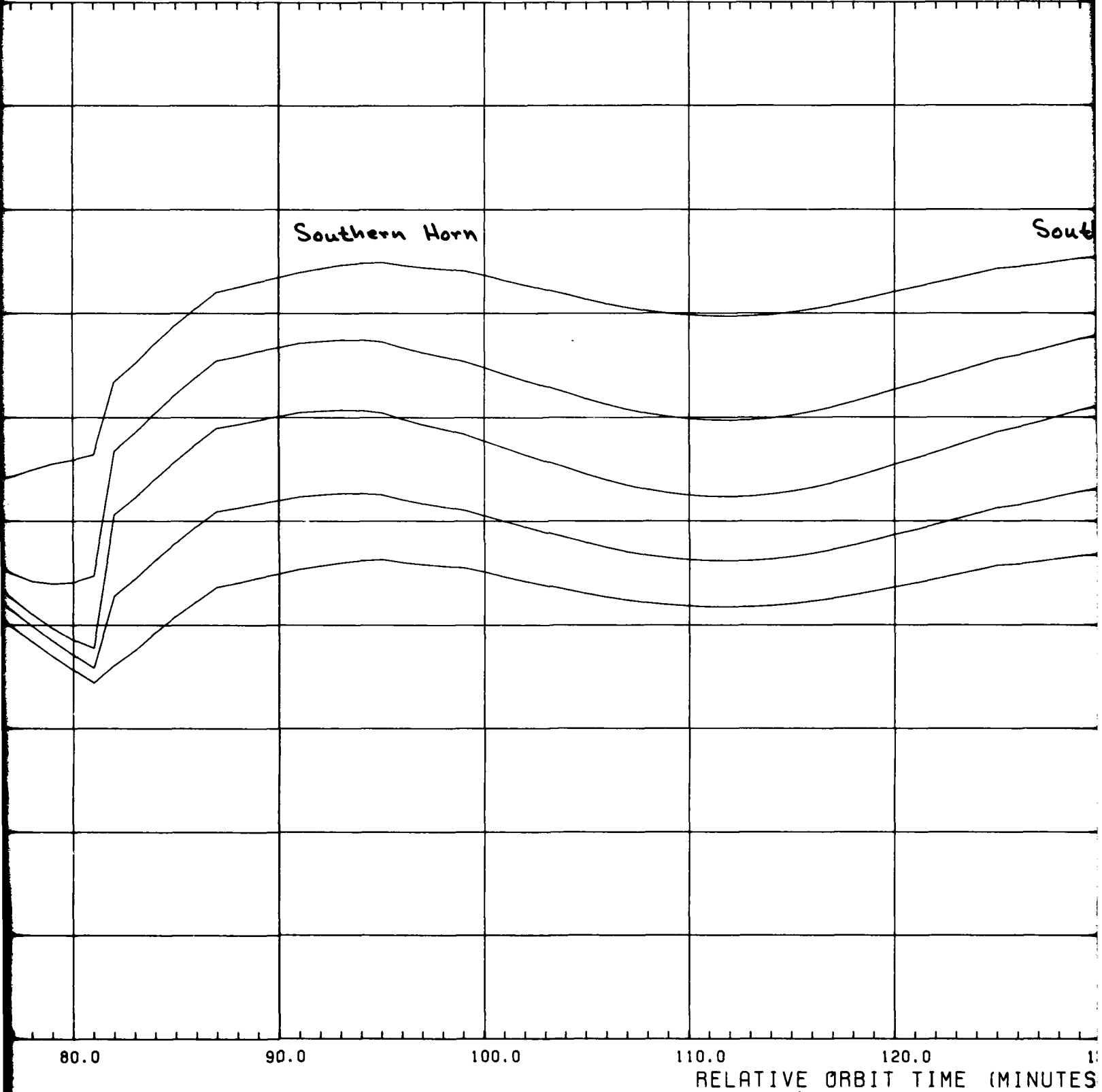
70.0

80.0



3

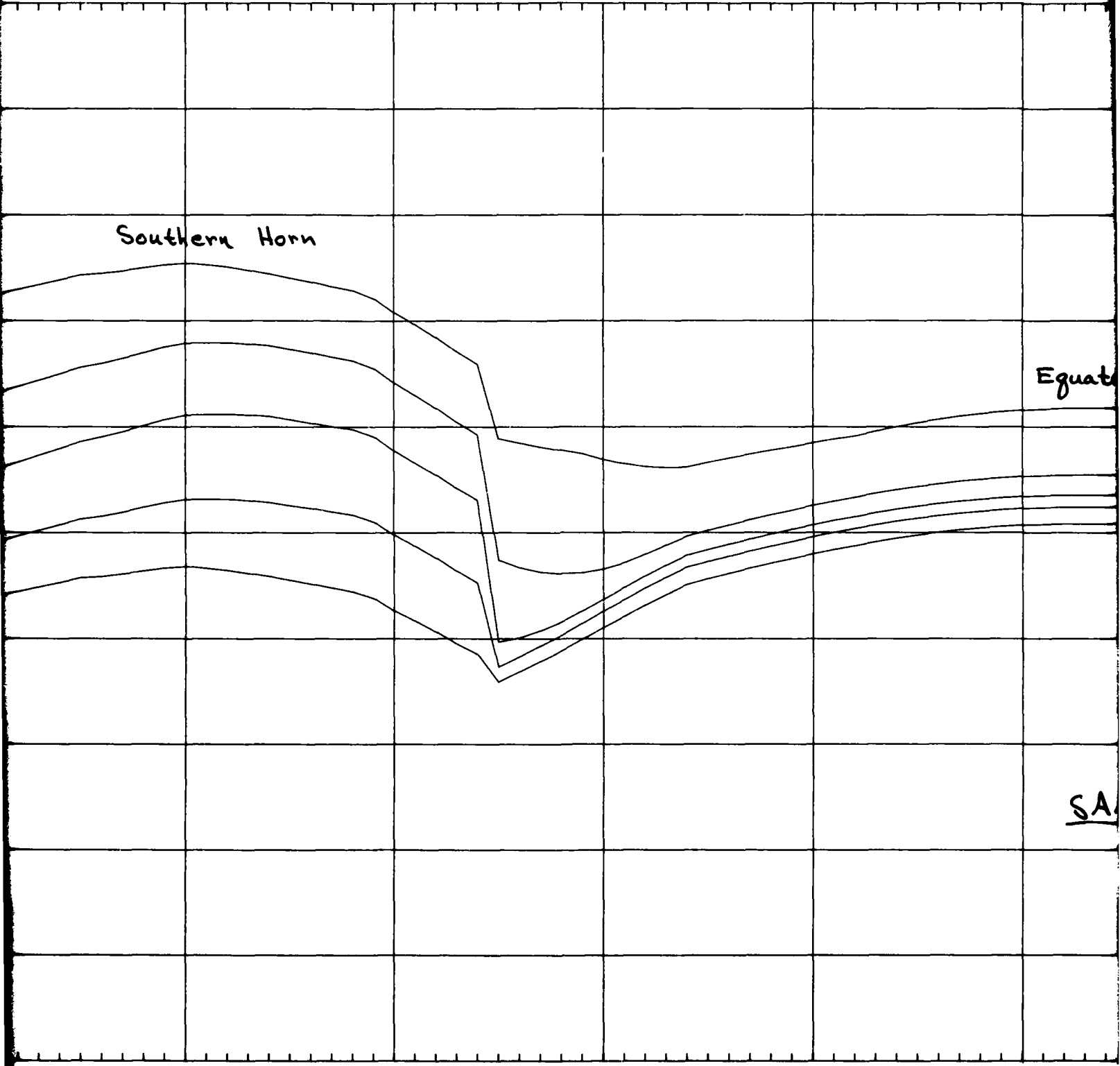
DOSE IN SEMI-INFINITE ALUMINUM



RELATIVE ORBIT TIME (MINUTES)

4

ITE ALUMINUM MEDIUM



Southern Horn

Equat

SA

TIME (MINUTES) 130.0 140.0 150.0 160.0 170.0

5

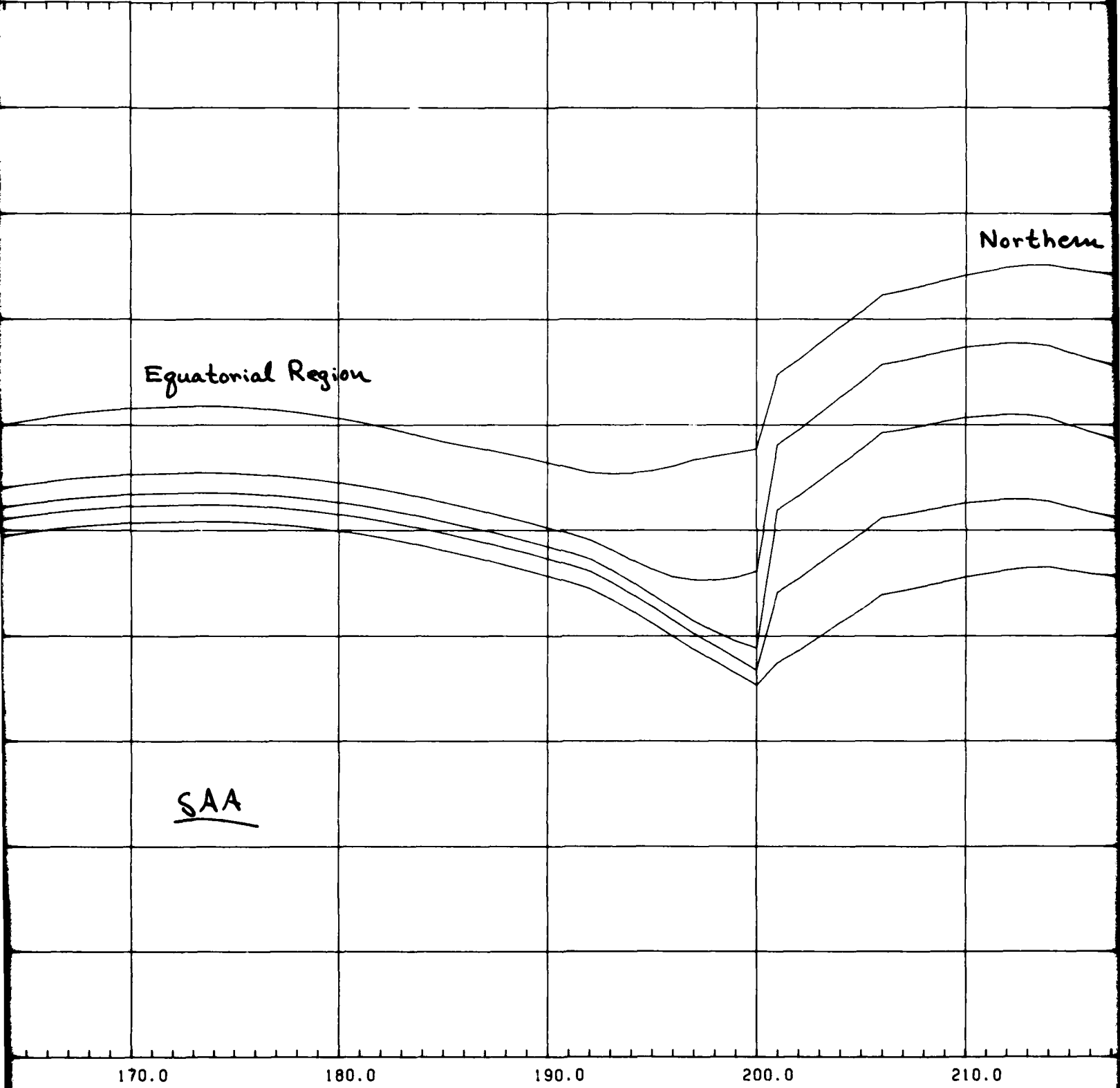


Figure 150

ORBIT: NAVELEX 5  
60 DGR/6389-6389 KM

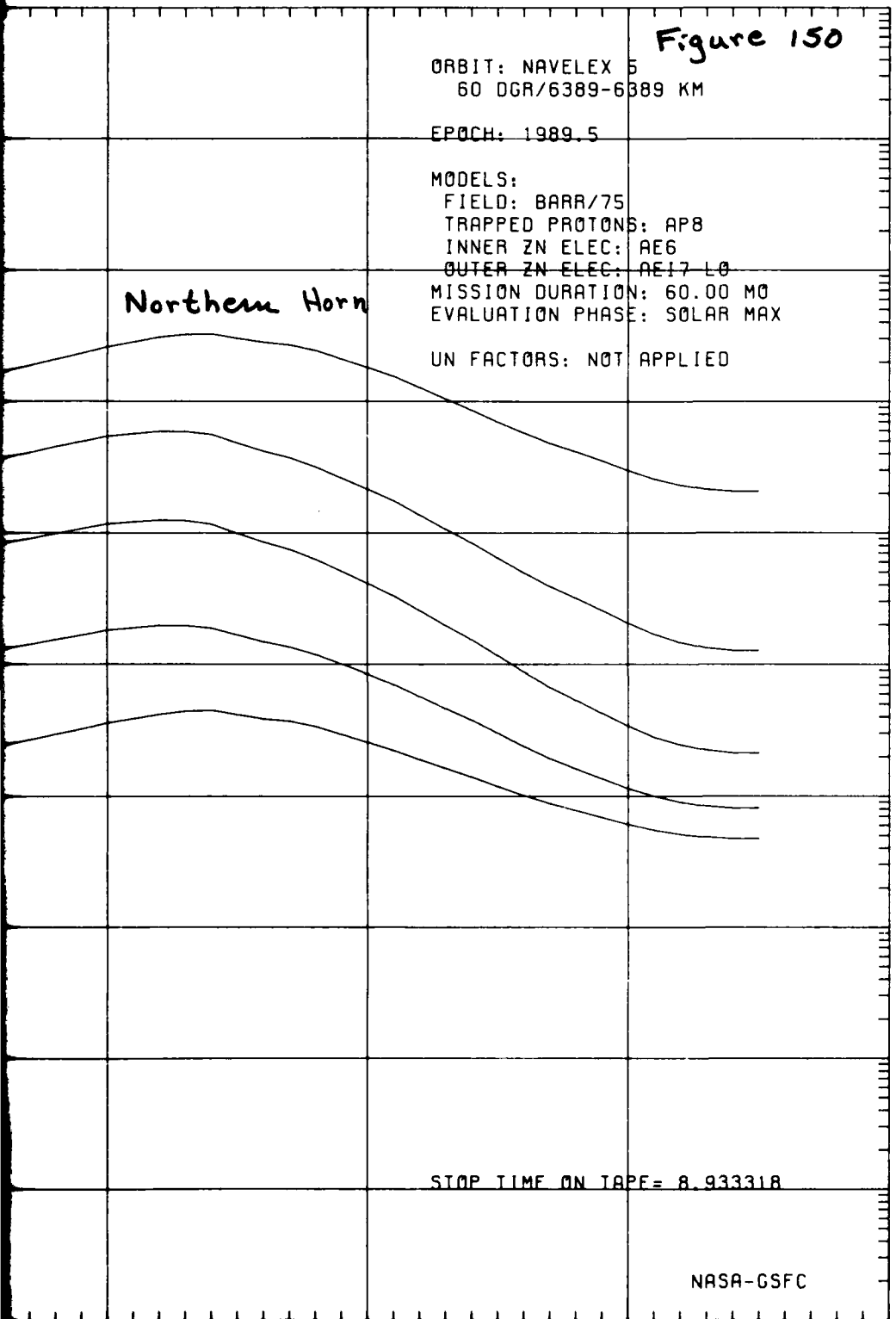
EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

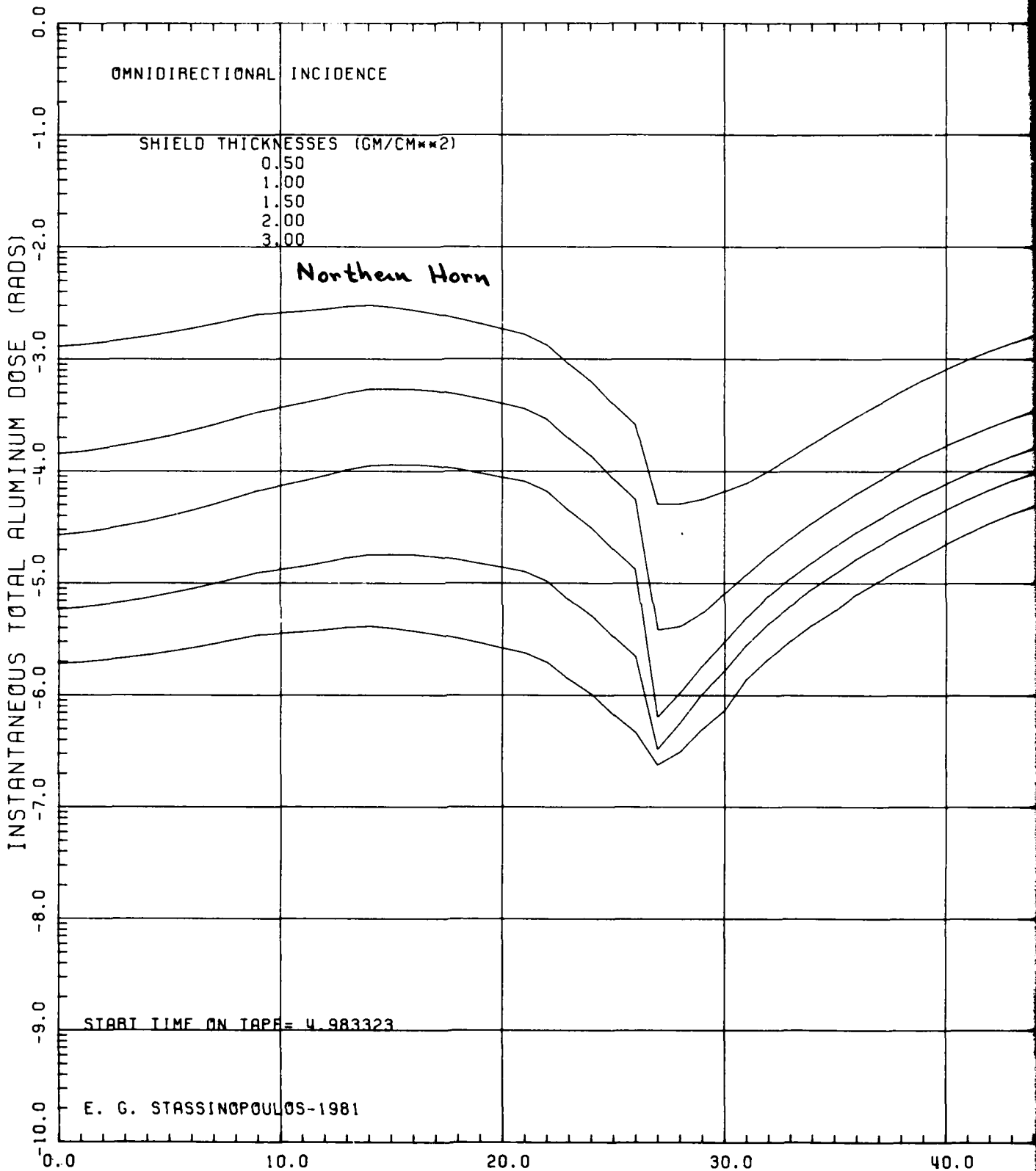
Northern Horn



STOP TIME ON TAPE = 8.933318

NASA-GSFC

210.0                      220.0                      230.0                      240.0





2'

Equatorial Region

SAA

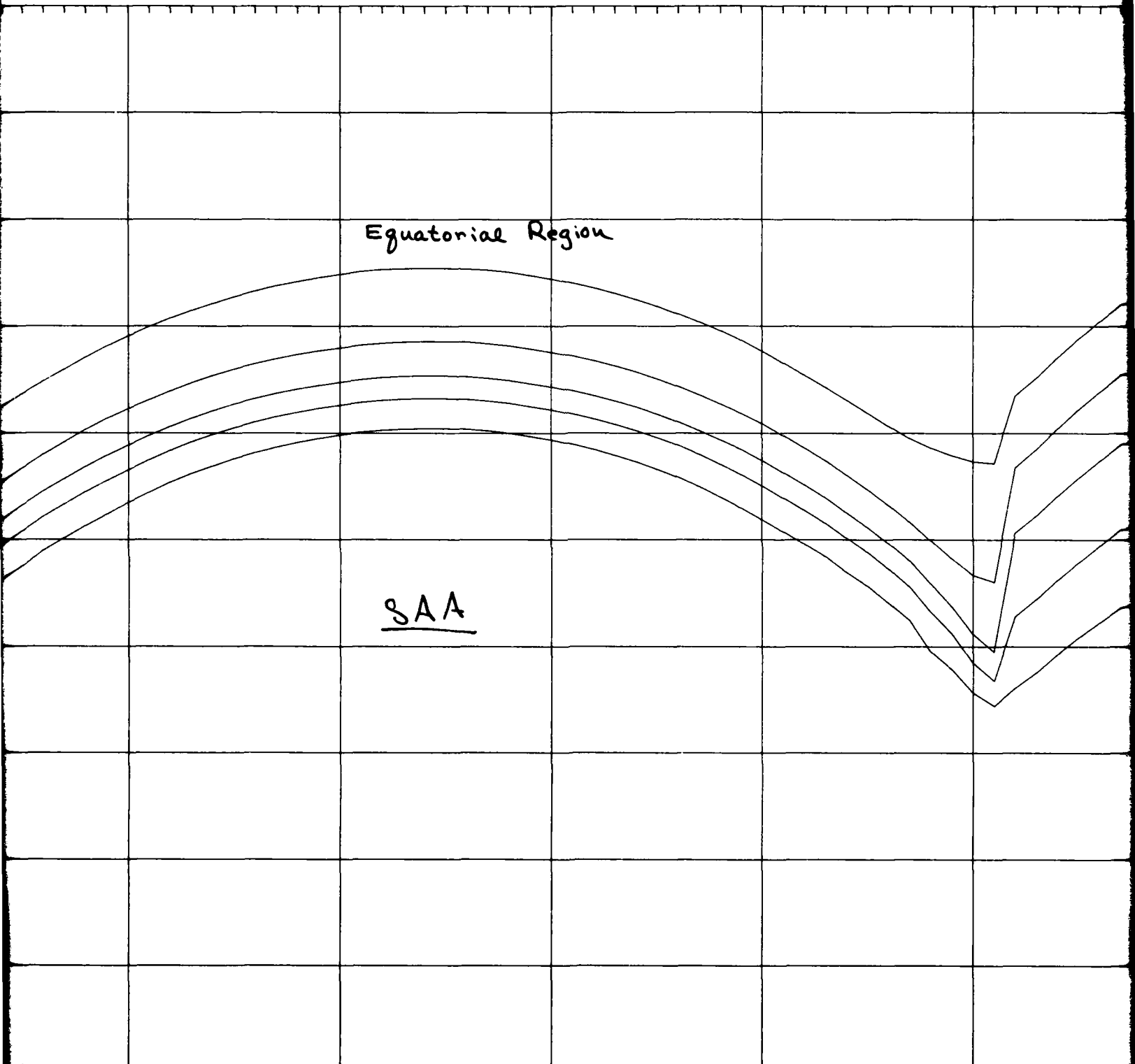
40.0

50.0

60.0

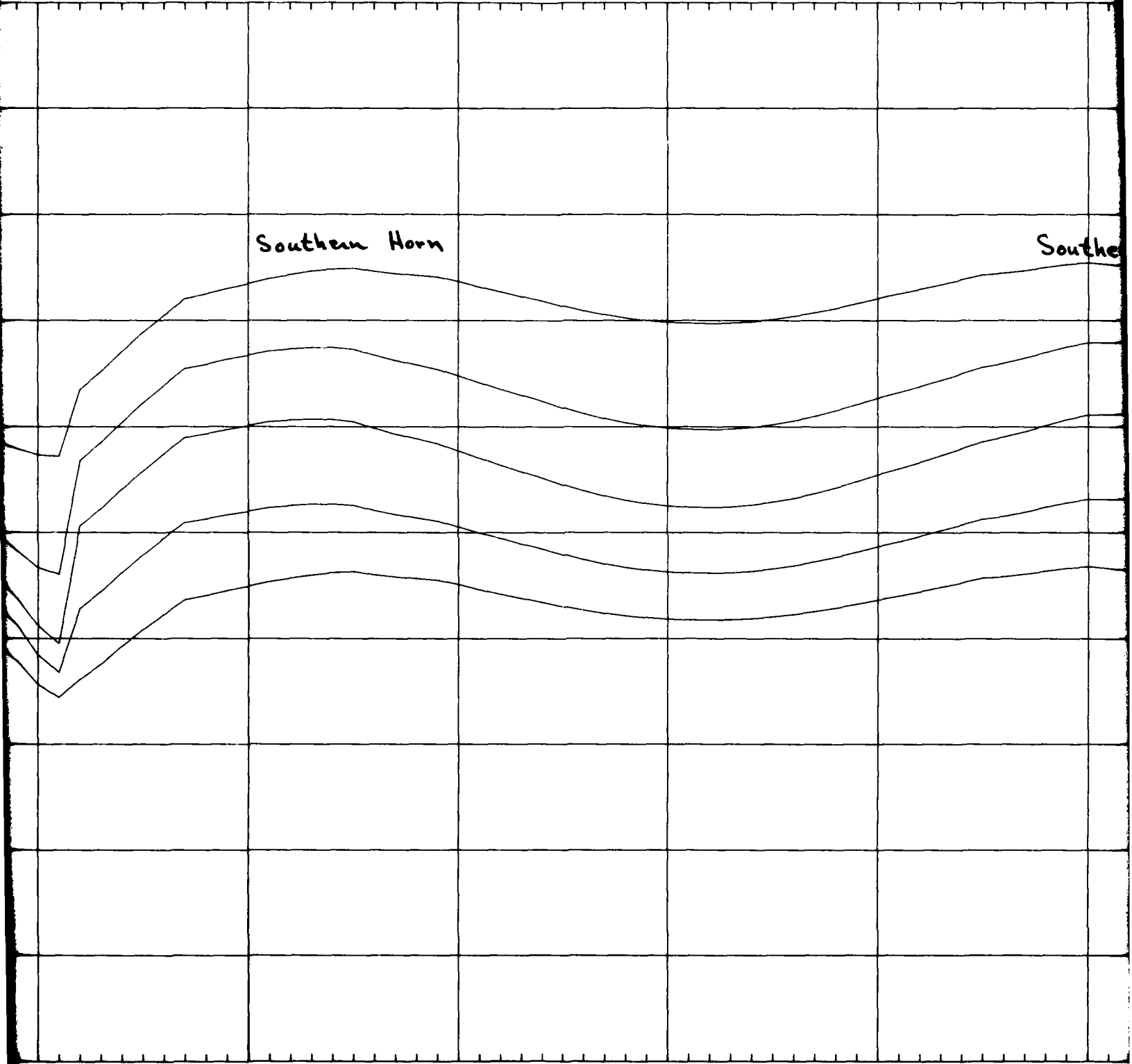
70.0

80.0



3

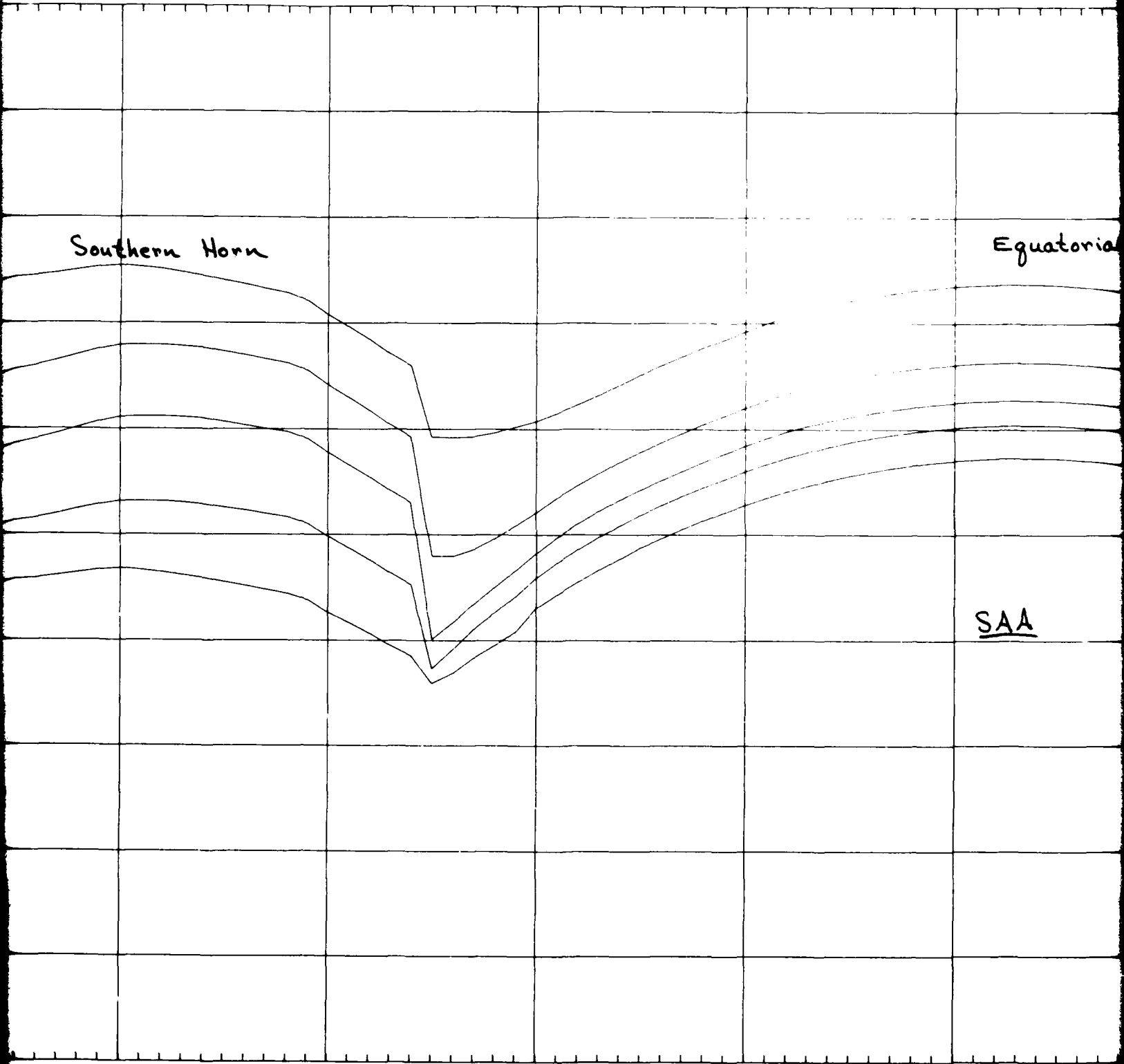
DOSE IN SEMI-INFINITE ALUMINUM M



80.0 90.0 100.0 110.0 120.0 130.0  
RELATIVE ORBIT TIME (MINUTES)

4

ALUMINUM MEDIUM



Southern Horn

Equatoria

SAA

(MINUTES)

140.0

150.0

160.0

170.0

51

Equatorial Region

Northern Horn

SAA

170.0 180.0 190.0 200.0 210.0 220.0

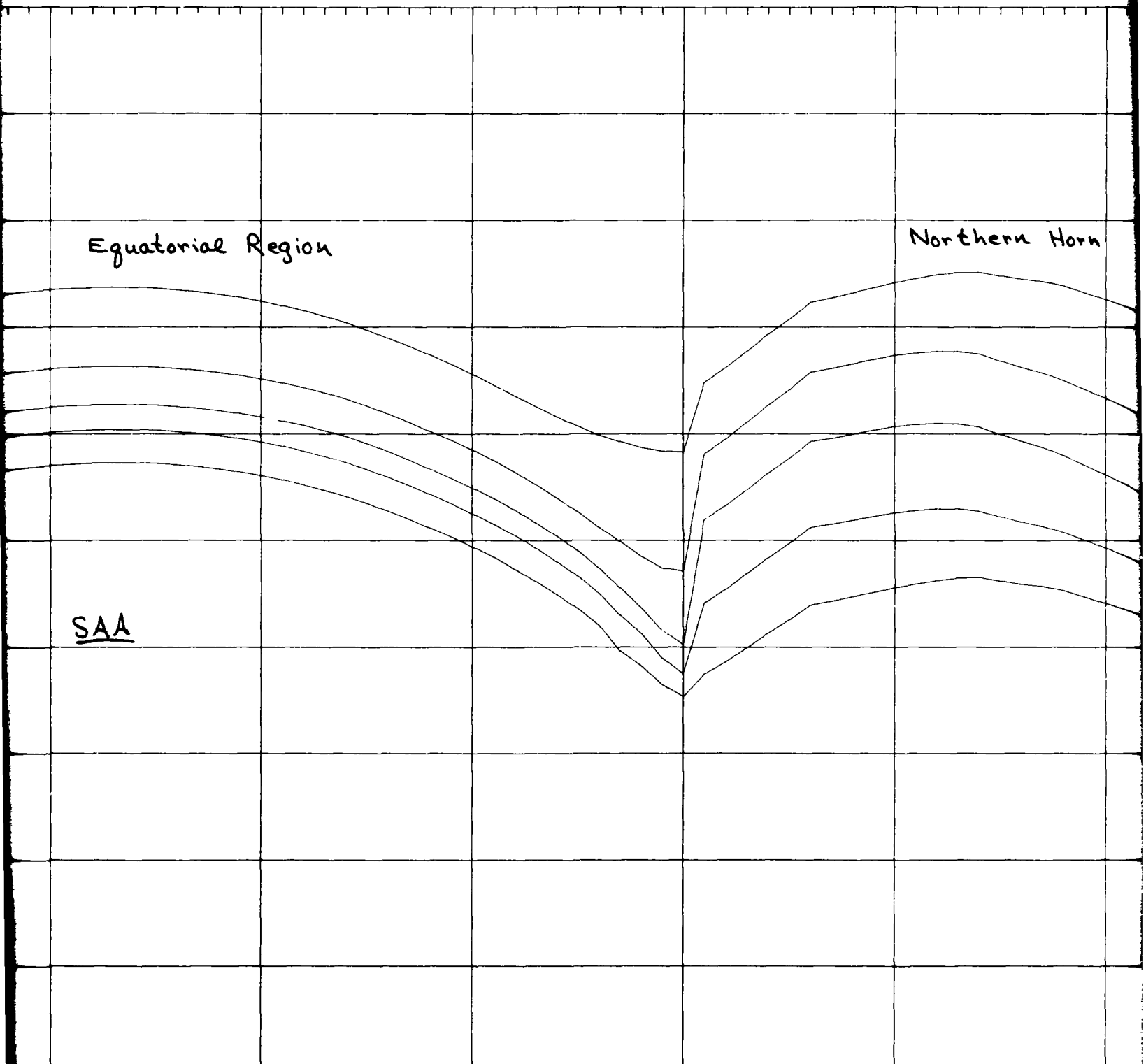


Figure 151

ORBIT: NAVELEX 5  
60 DGR/6389-6389 KM

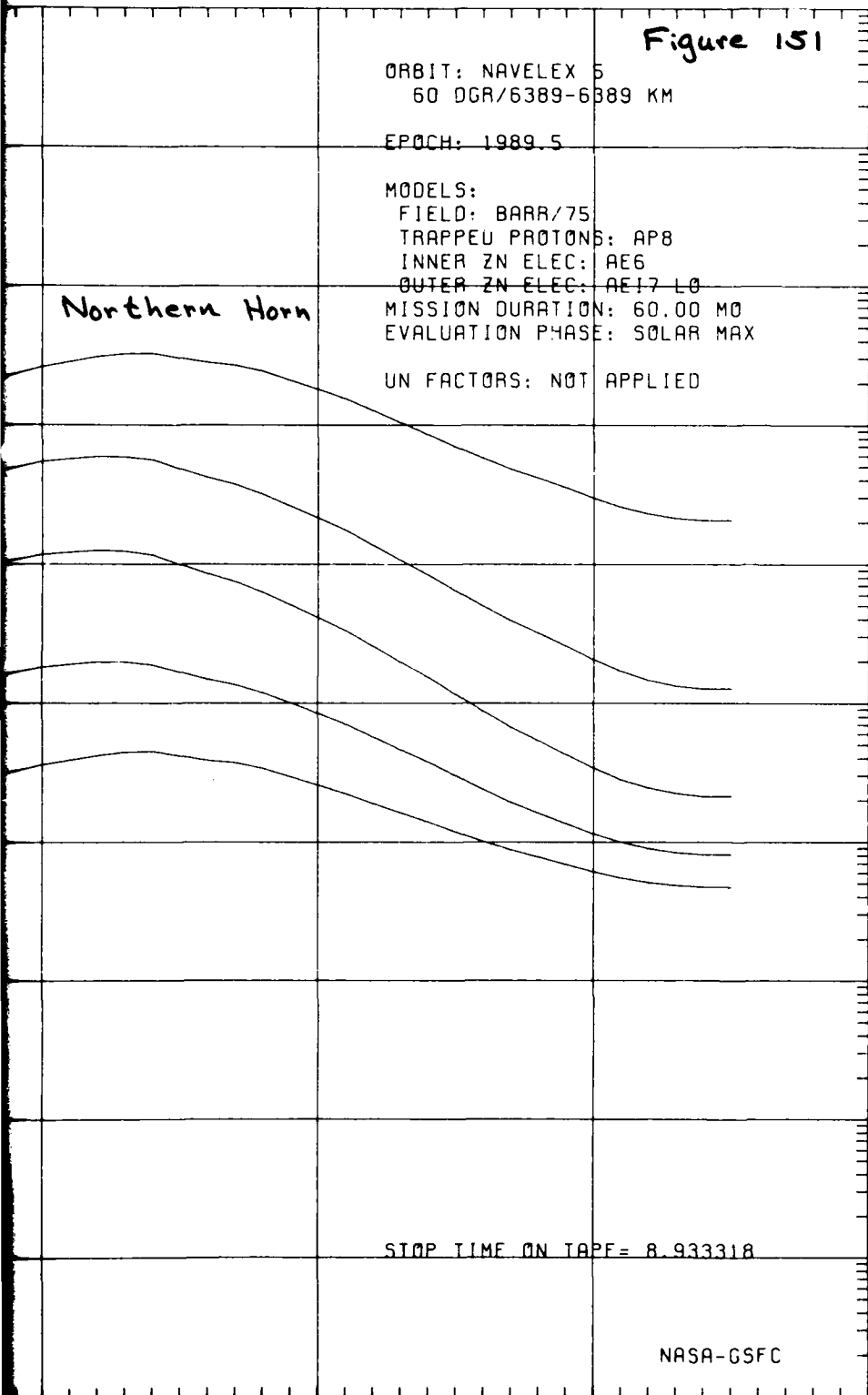
EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Northern Horn



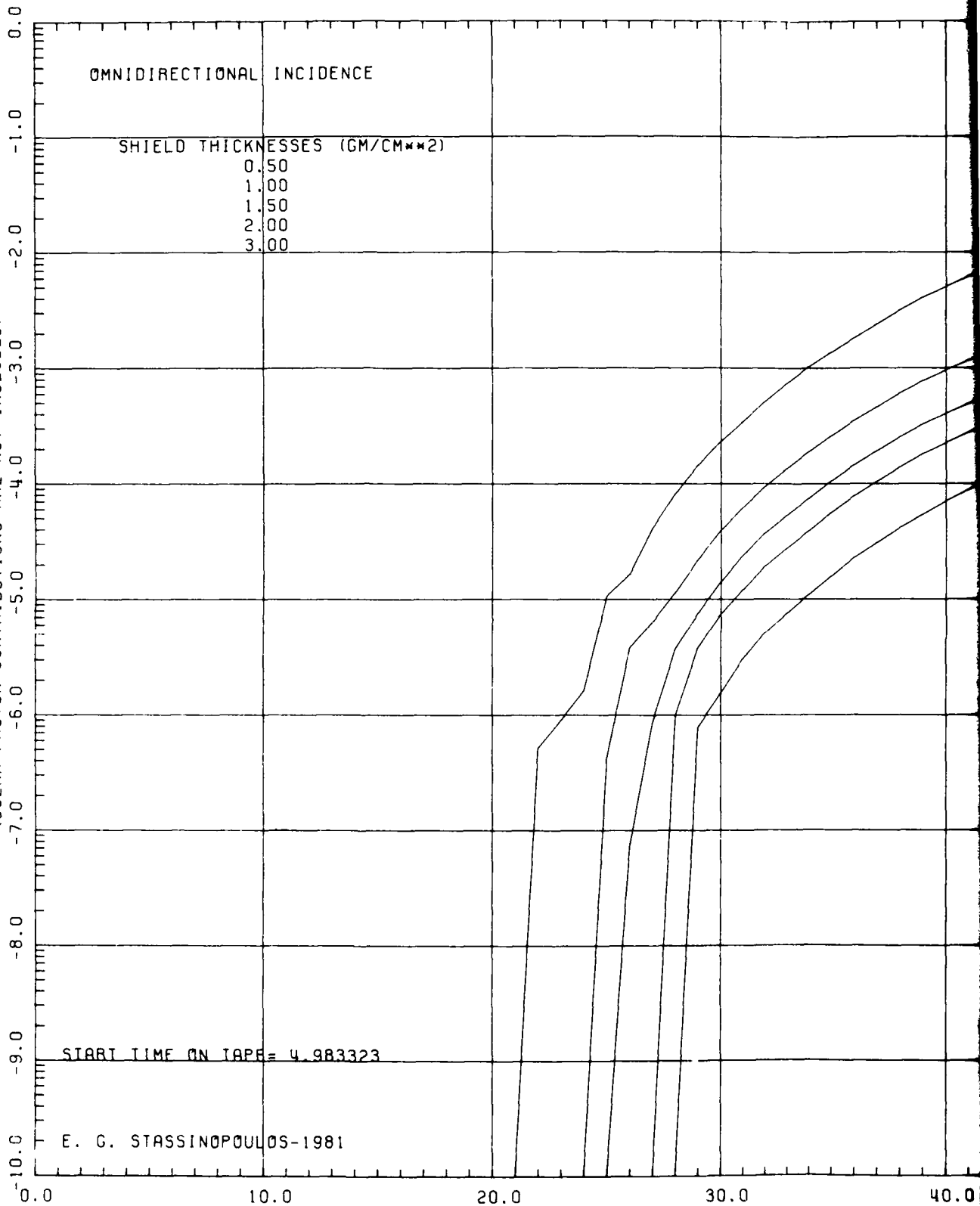
STOP TIME ON TAPE = 8.933318

NASA-GSFC

210.0 220.0 230.0 240.0

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



1  
2

Equatorial Region

SAA

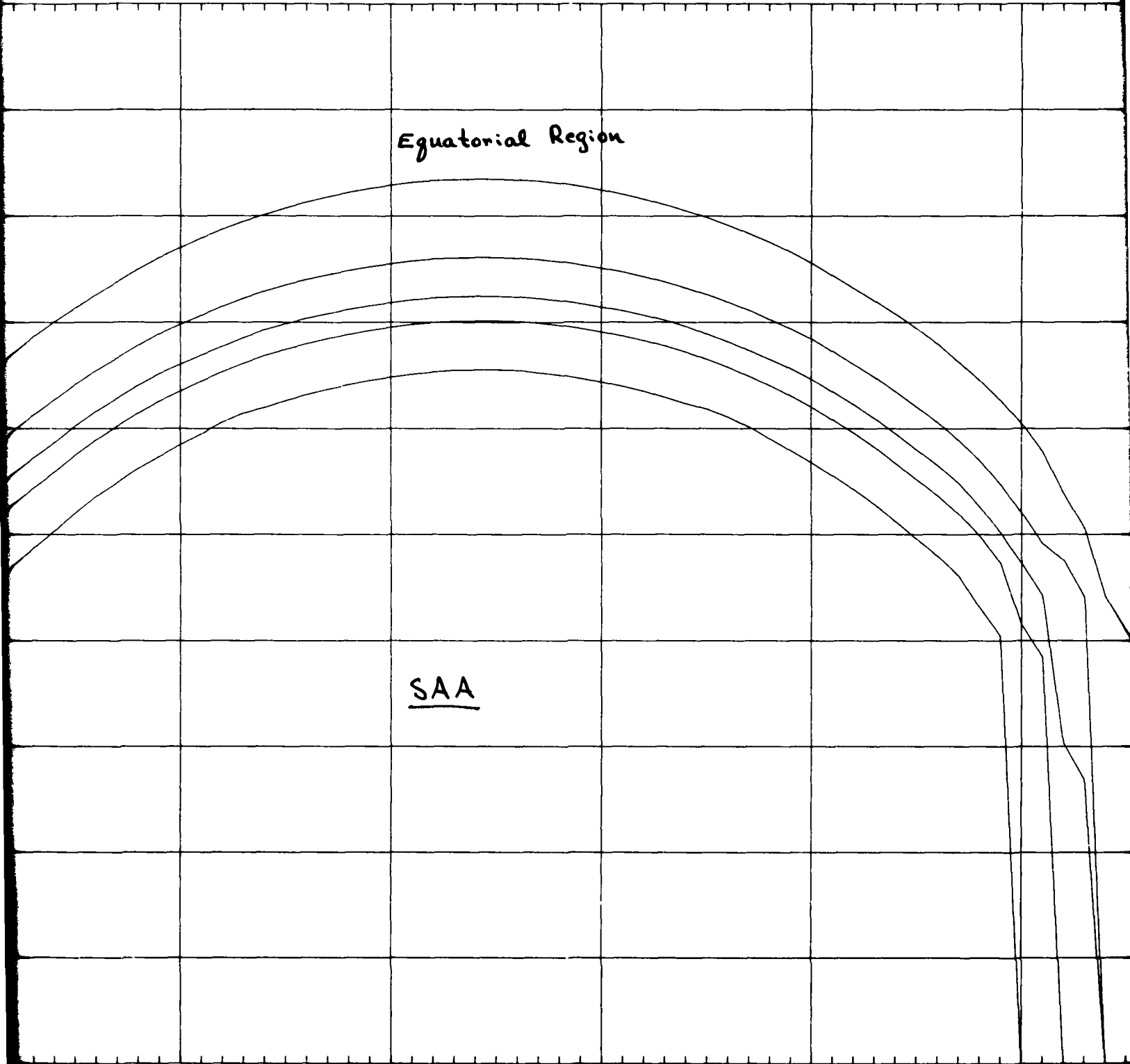
40.0

50.0

60.0

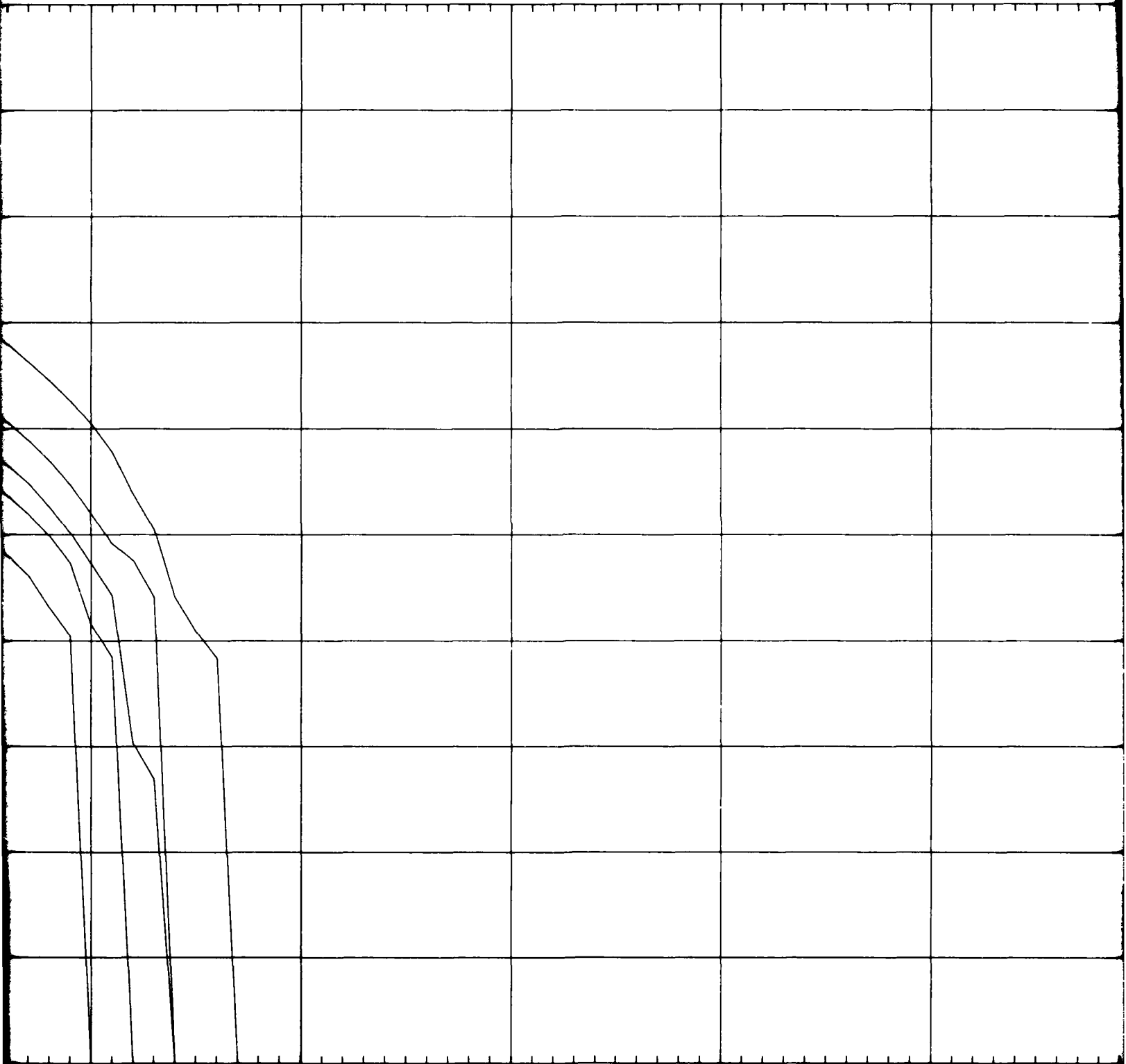
70.0

80.0



3

DOSE AT CENTER OF ALUMINUM



80.0

90.0

100.0

110.0

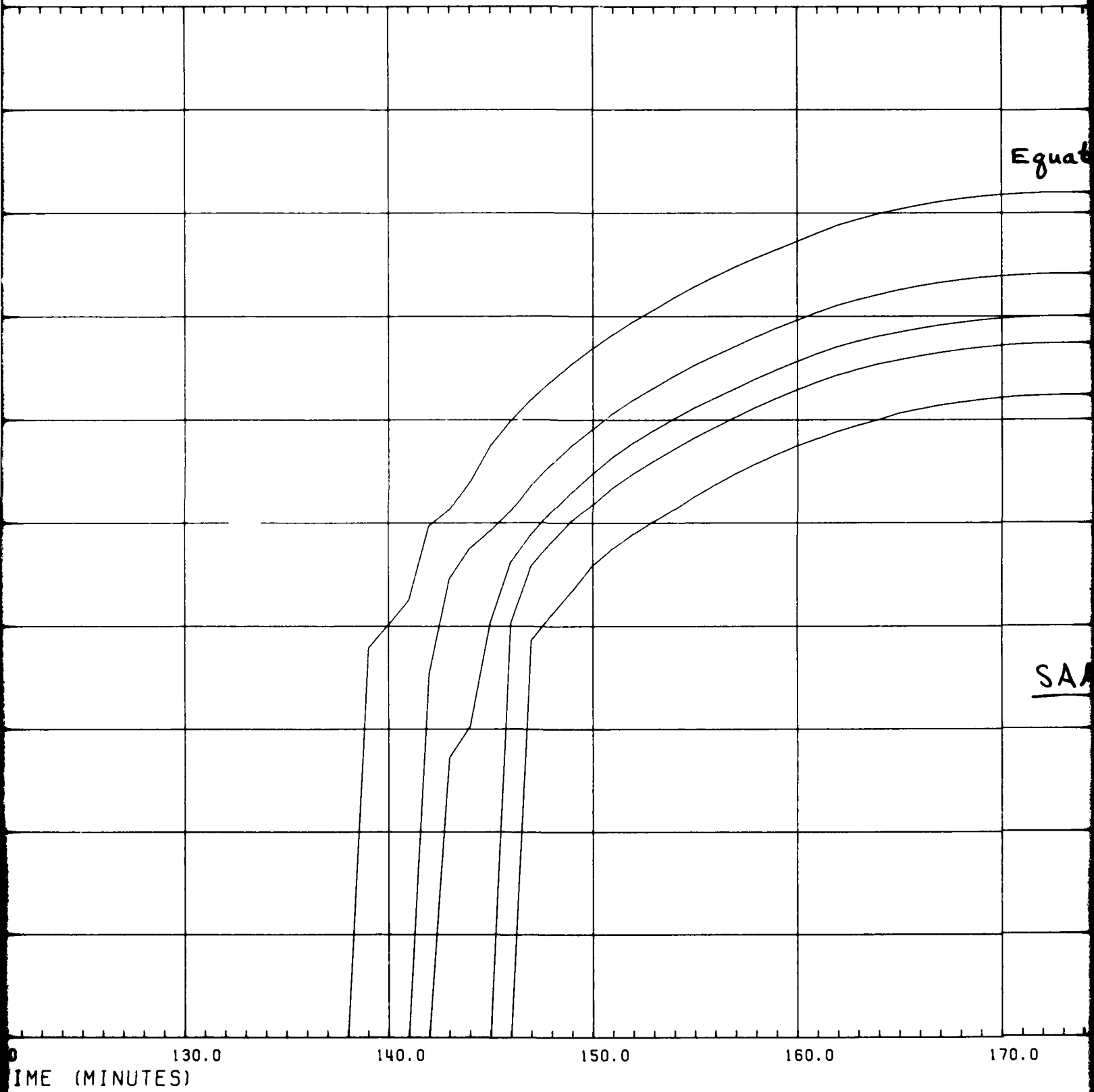
120.0

RELATIVE ORBIT TIME (MINUTE)



4

ALUMINUM SPHERES

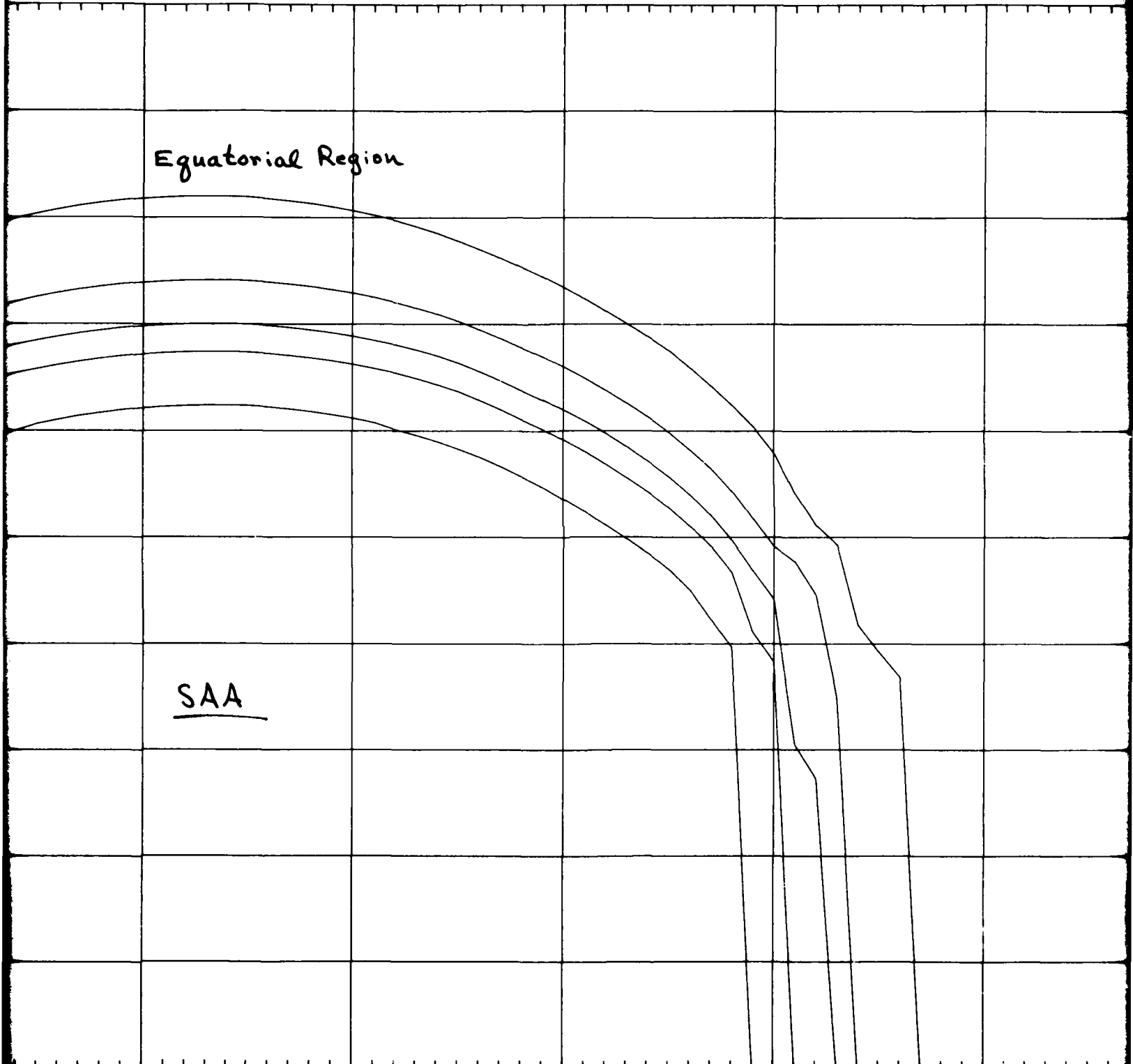


5

Equatorial Region

SAA

170.0 180.0 190.0 200.0 210.0



6

Figure 15a

ORBIT: NAVELEX 5  
60 DGR/6389-6389 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX  
  
UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 8.933318

NASA-GSFC

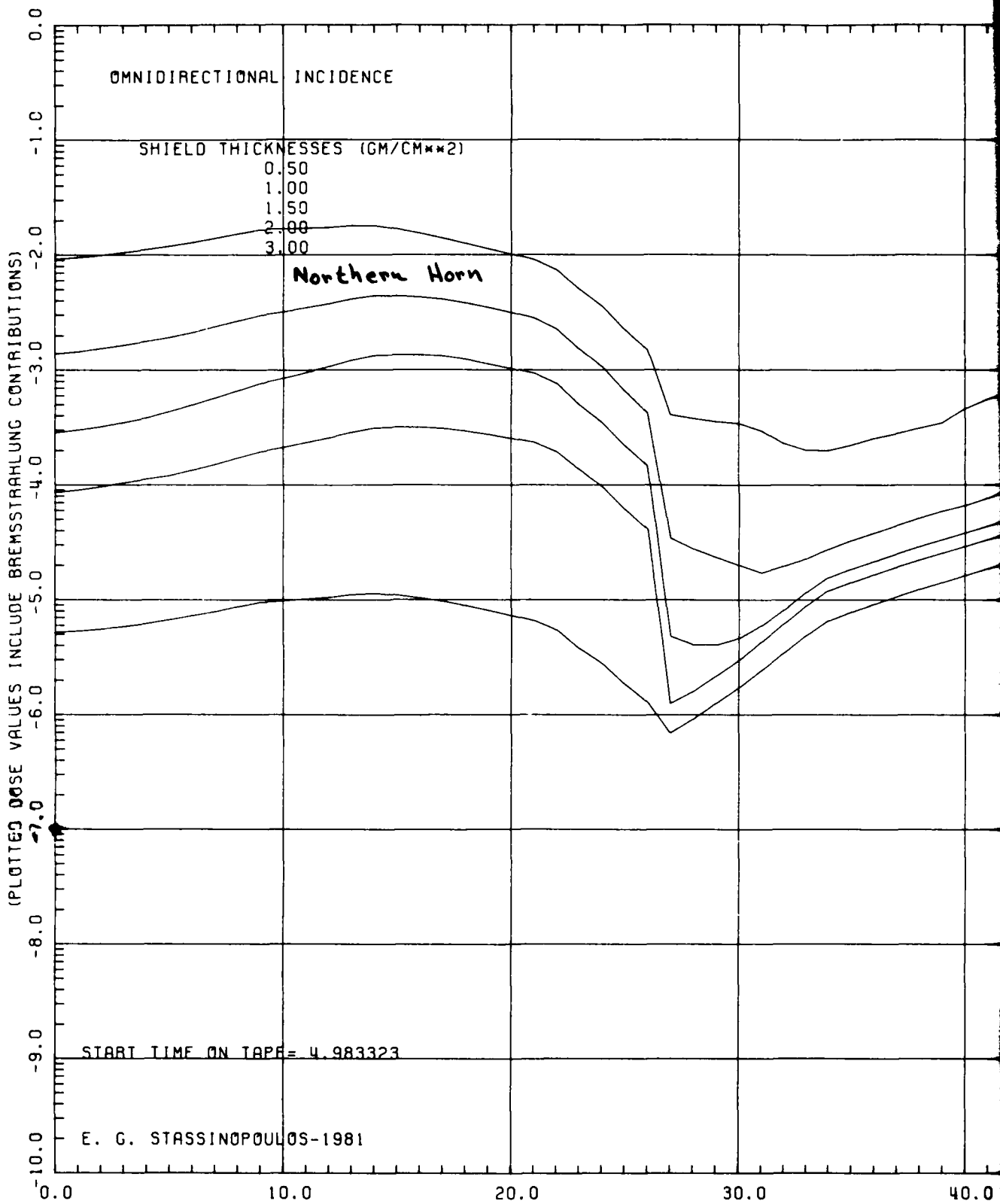
210.0

220.0

230.0

240.0

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)



2<sup>1</sup>

Equatorial Region

SAA

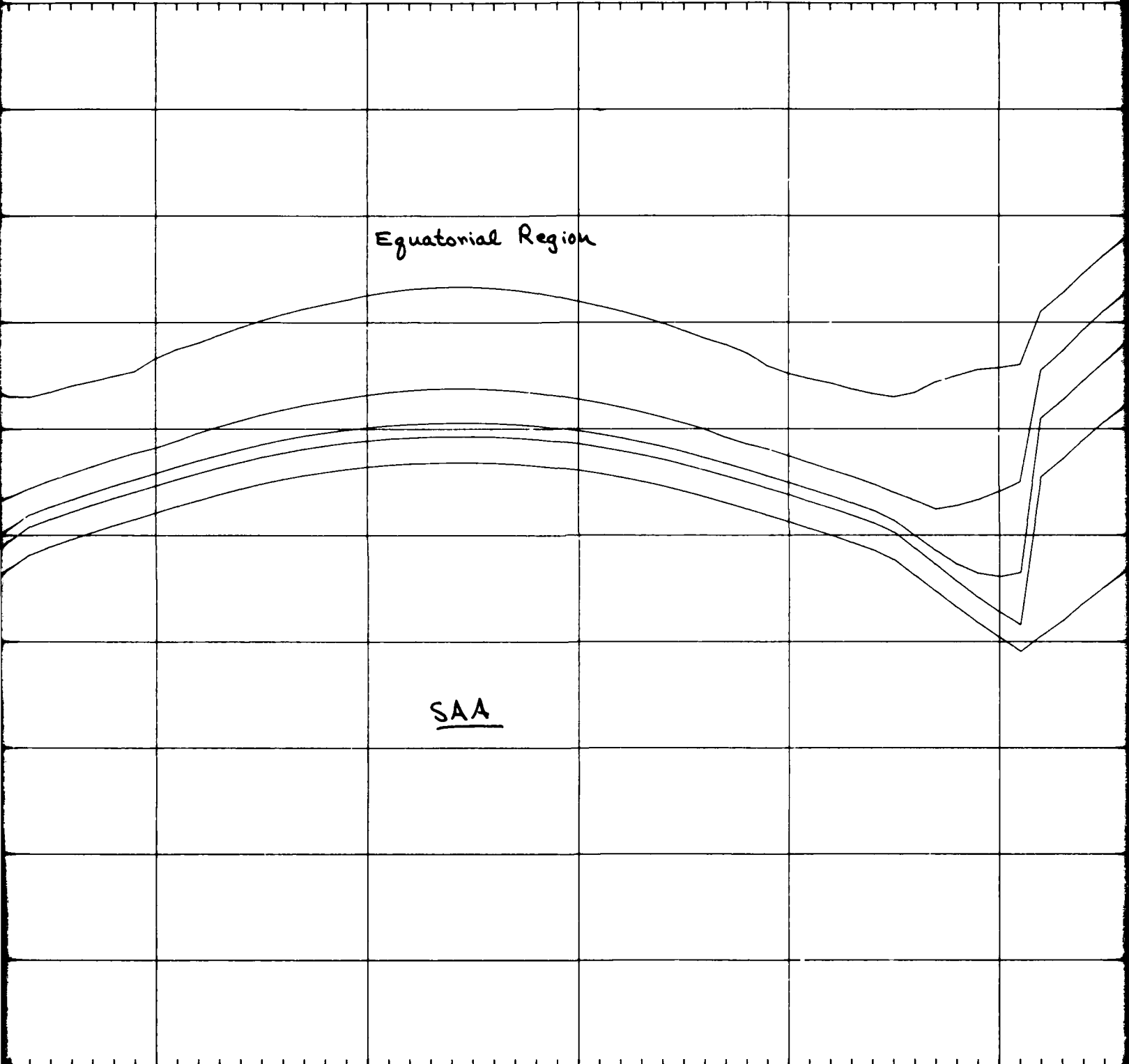
40.0

50.0

60.0

70.0

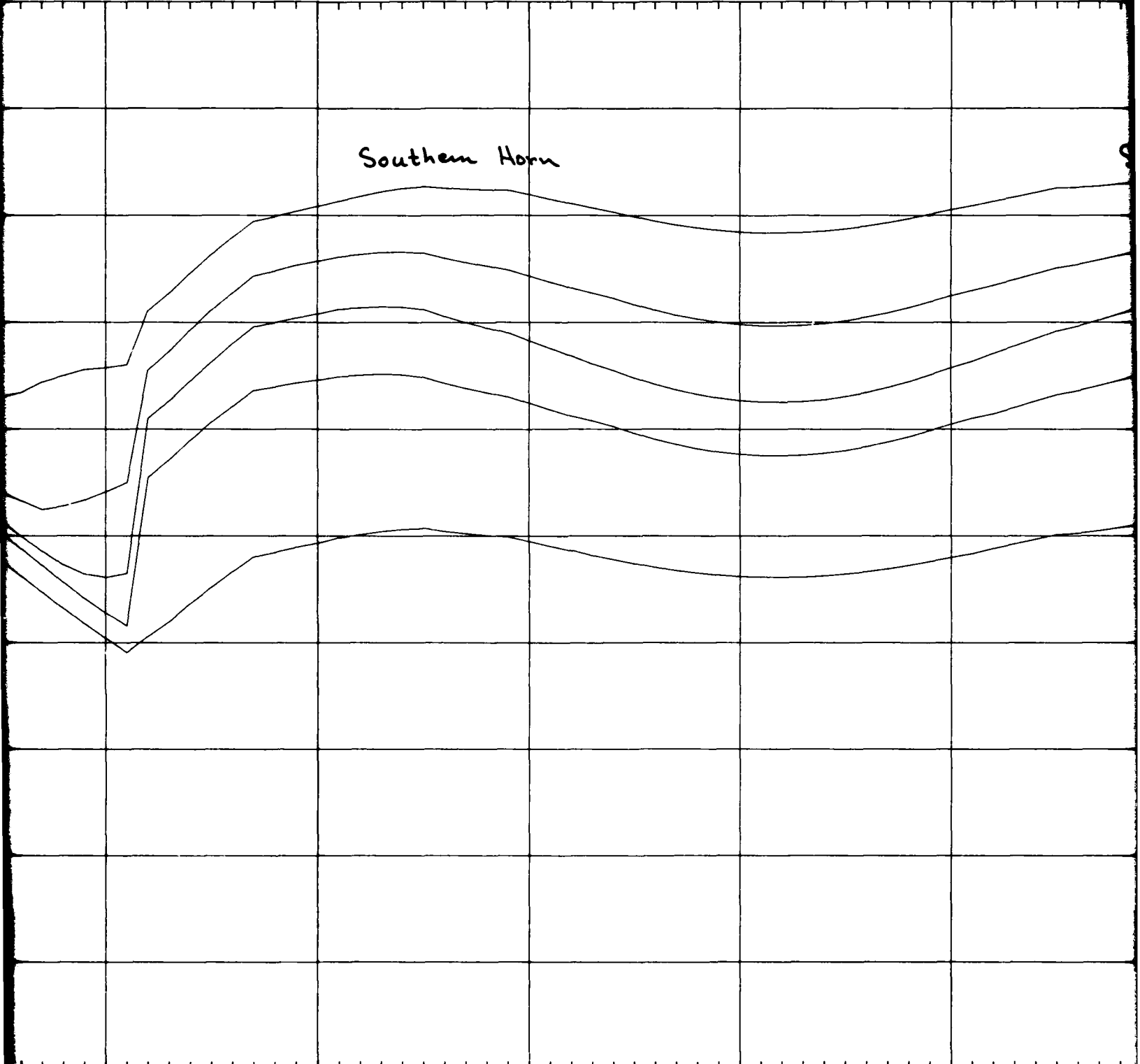
80.0



3<sup>1</sup>

DOSE AT CENTER OF ALUMINUM

Southern Horn



80.0

90.0

100.0

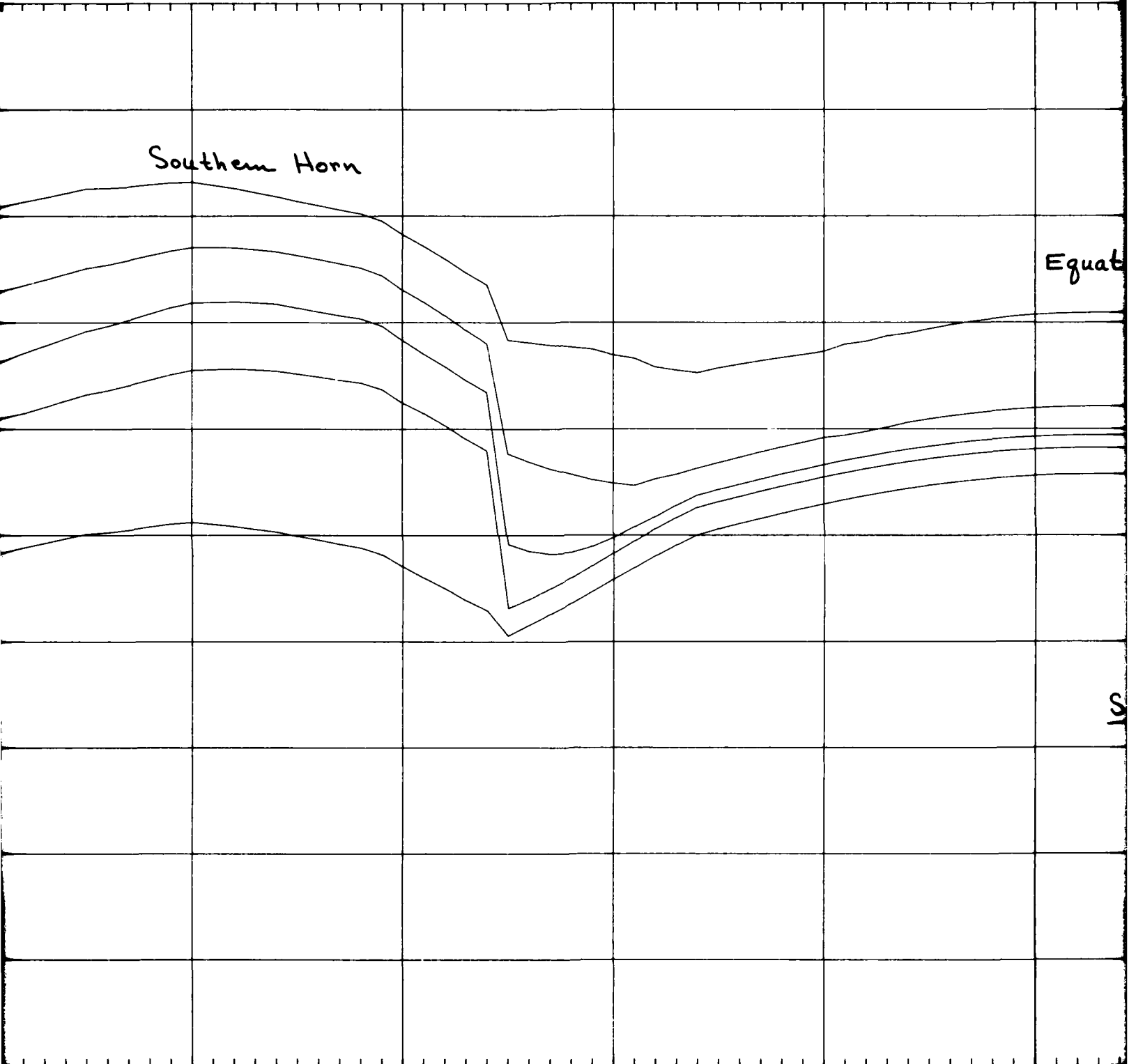
110.0

120.0

RELATIVE ORBIT TIME (MINUT)

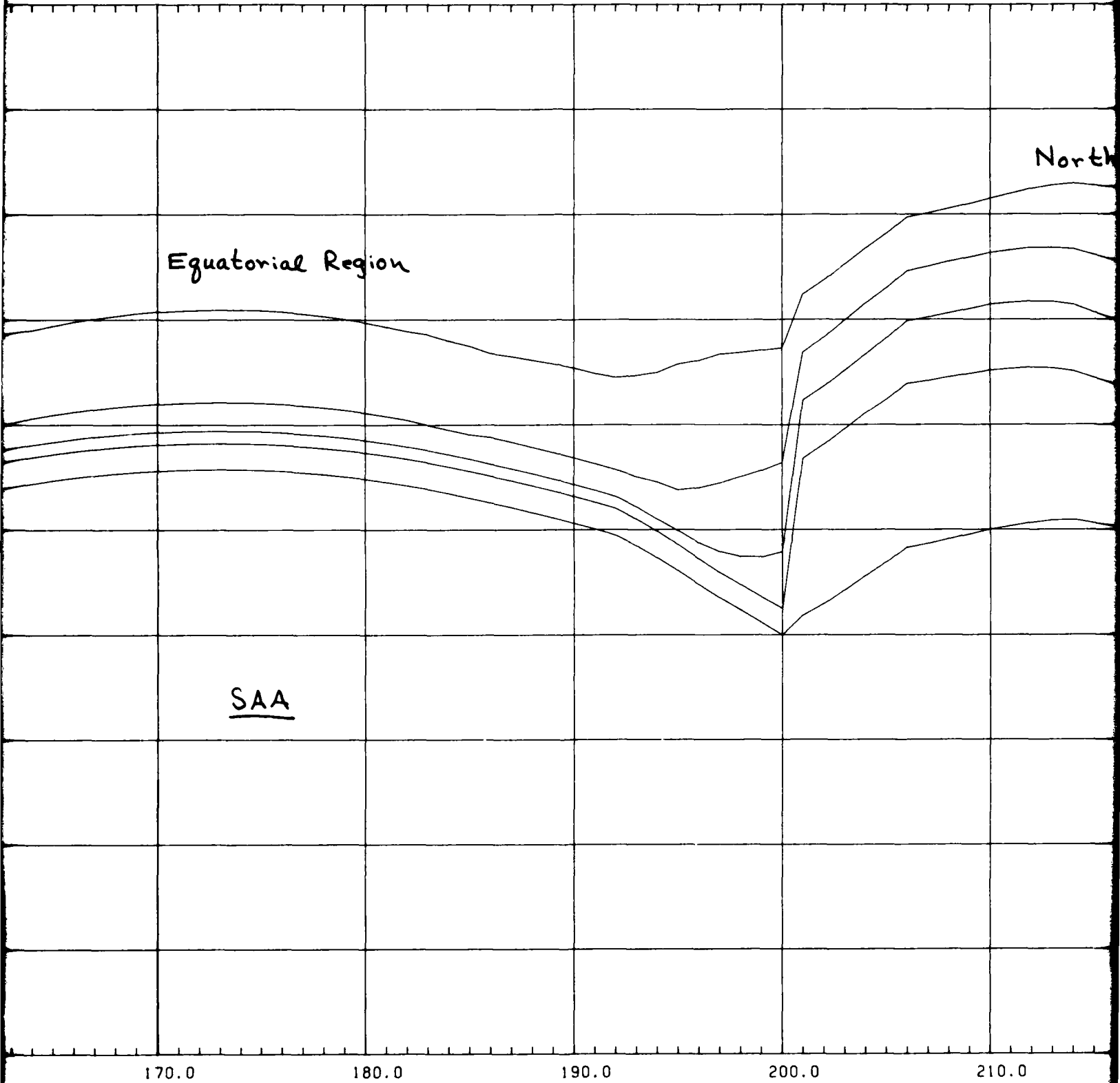
4

F ALUMINUM SPHERES



0 130.0 140.0 150.0 160.0 170.0  
TIME (MINUTES)

5





6

Figure 153

ORBIT: NAVELEX 5  
60 DGR/6389-6389 KM

EPOCH: 1989.5

Northern  
Horn

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPF = 8.933318

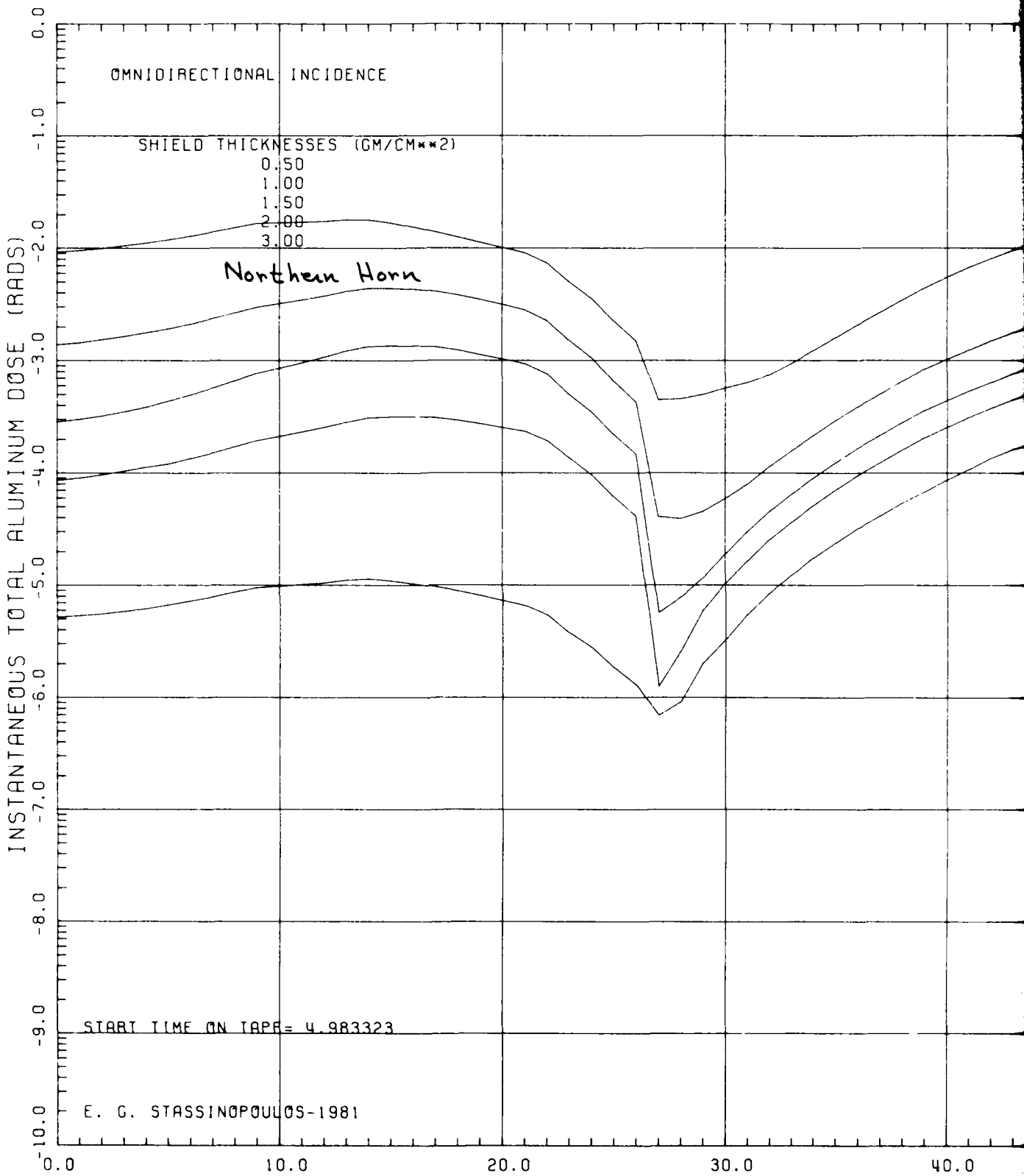
NASA-GSFC

210.0

220.0

230.0

240.0



Equatorial Region

SAA

40.0

50.0

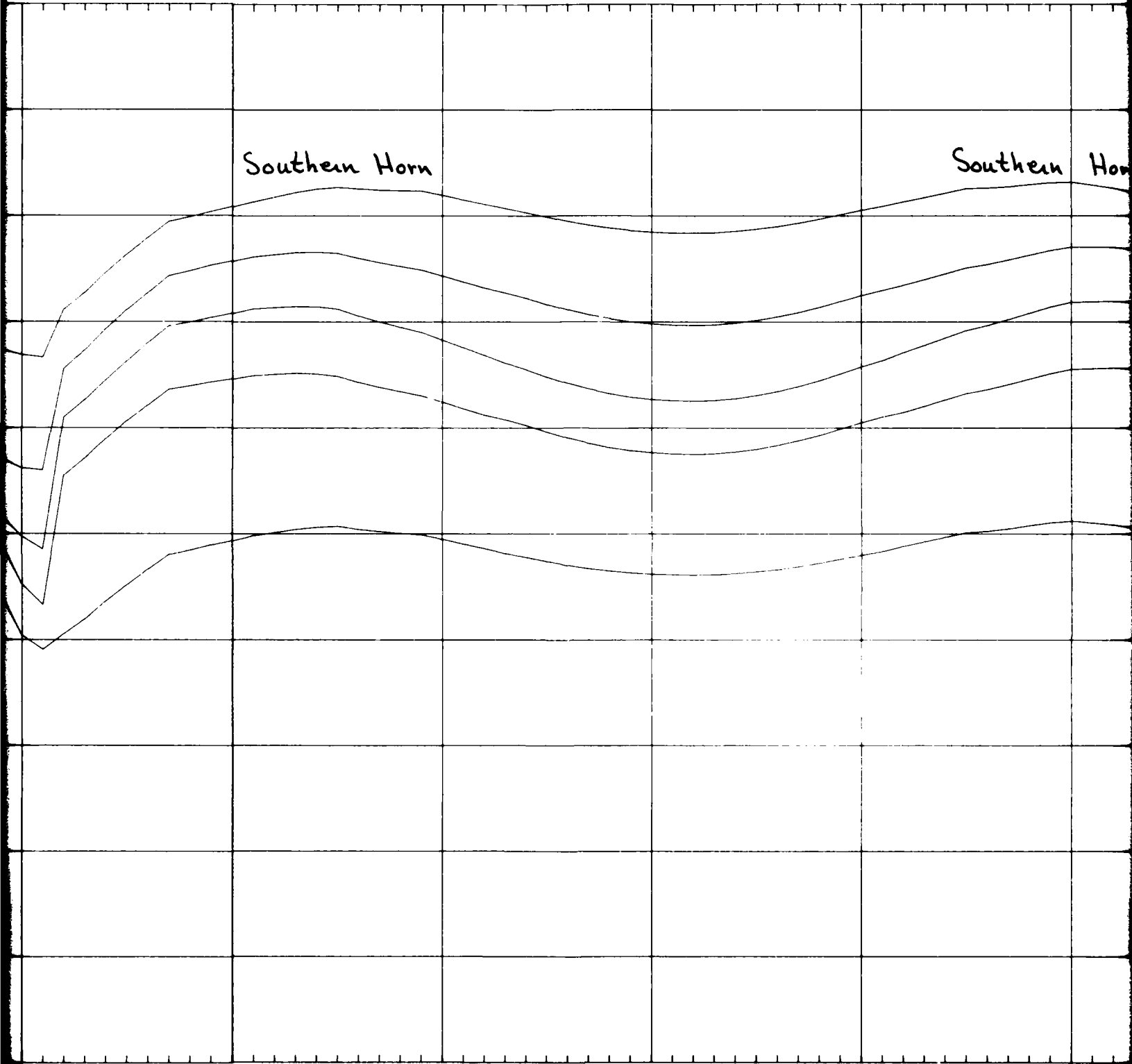
60.0

70.0

80.0

3

DOSE AT CENTER OF ALUMINUM SPHERE



Southern Horn

Southern Horn

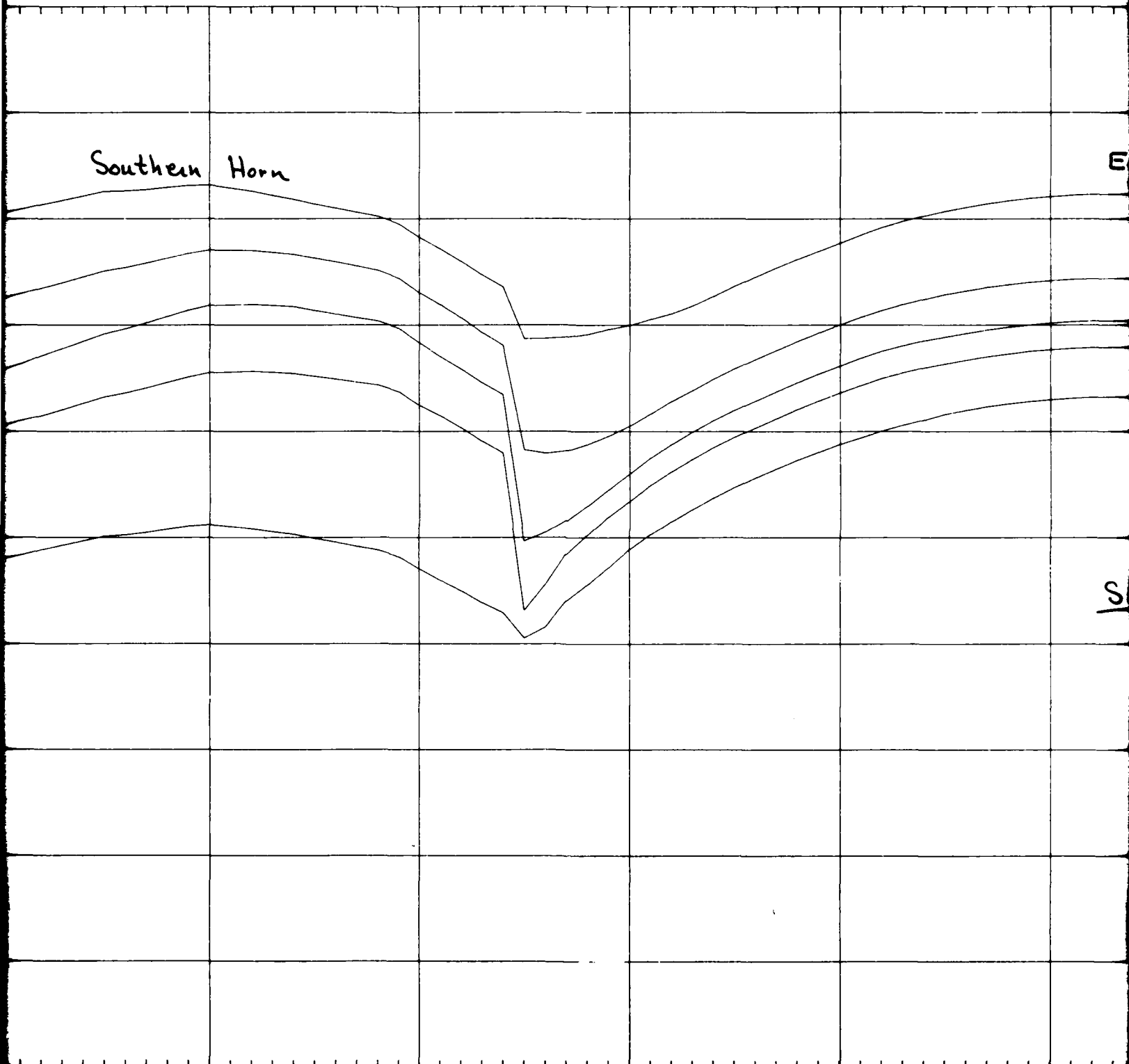
0.0 90.0 100.0 110.0 120.0 130.0

RELATIVE ORBIT TIME (MINUTES)

4'

OF ALUMINUM SPHERES

Southern Horn



0.0 130.0 140.0 150.0 160.0 170.0  
TIME (MINUTES)

5

Equatorial Region

Northern

SAA

170.0

180.0

190.0

200.0

210.0

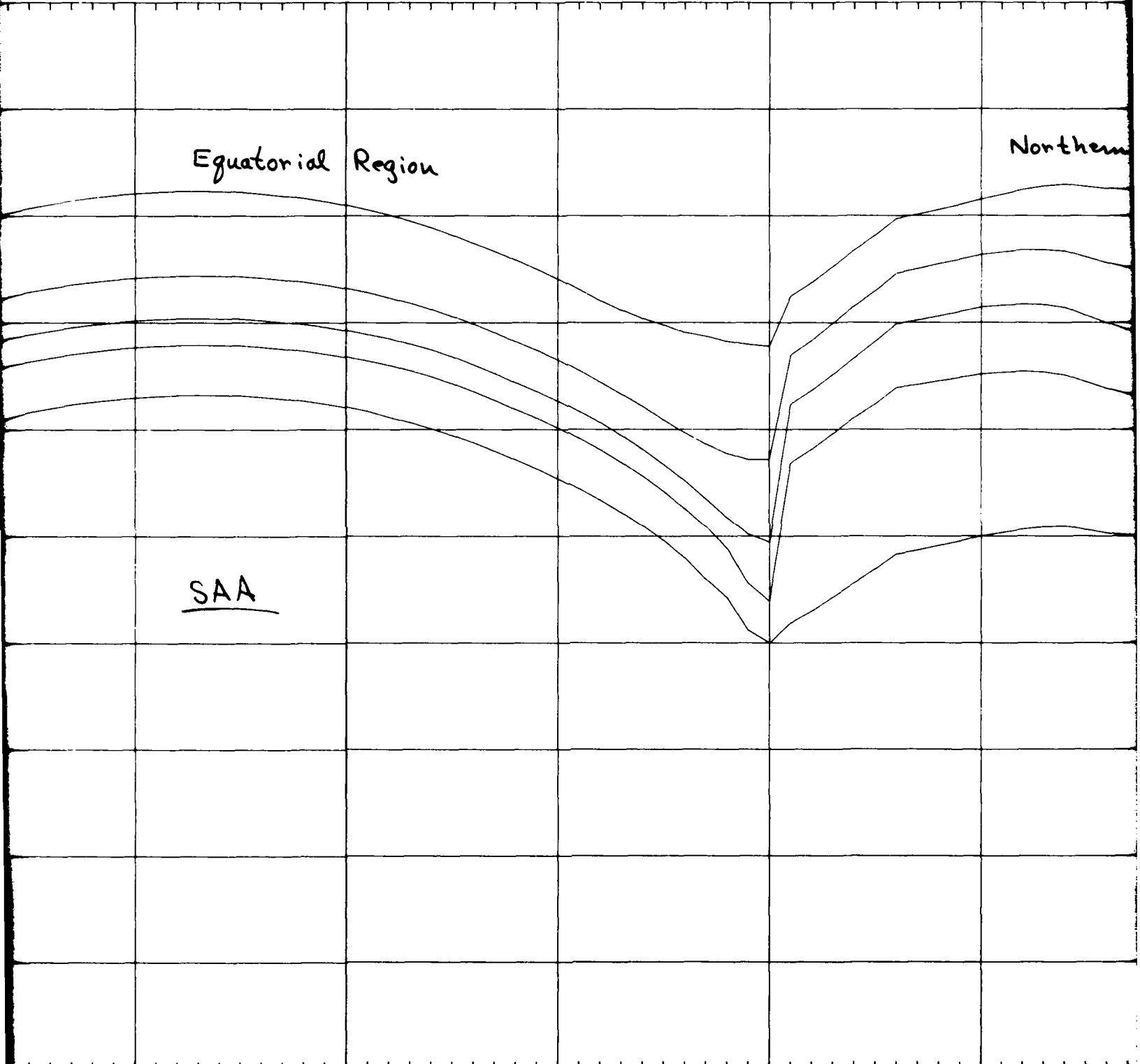


Figure 154

Northern Horn

ORBIT: NAVELEX 5  
60 DGR/6389-6389 KM

EPOCH: 1989.5

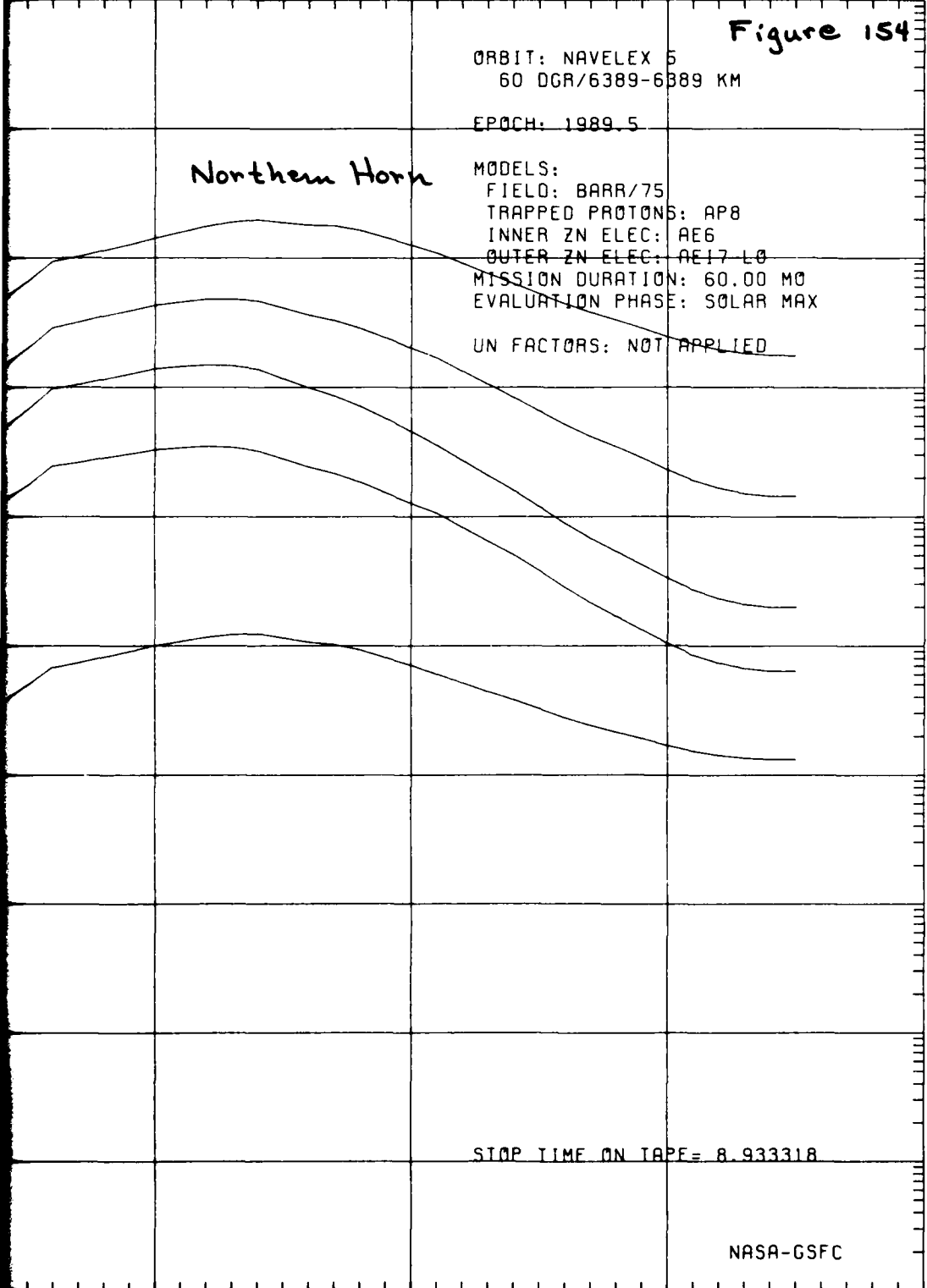
MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AEG  
OUTER ZN ELEC: AE17 LB  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

STOP TIME ON TAPE = 8.933318

NASA-GSFC

210.0 220.0 230.0 240.0



DOSE AT TRANSMISSION SURFACE OF FINI

(= DOSE IN SEMI-INFINITE ALUMINUM ME

OMNIDIRECTIONAL INCIDENCE

SHIELD THICKNESSES (GM/CM\*\*2)

- 0.50
- 1.00
- 1.50
- 2.00
- 3.00

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

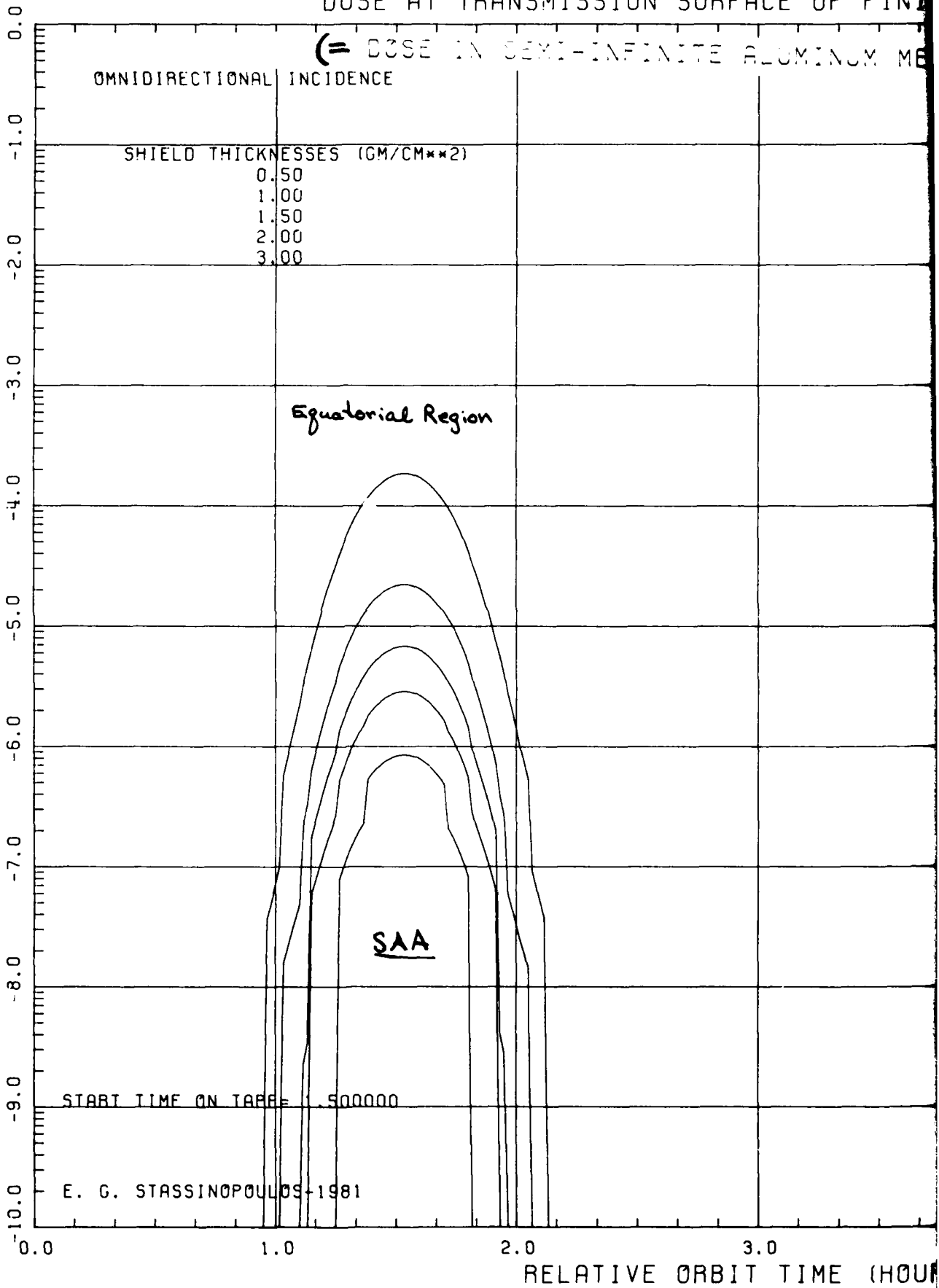
(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)

Equatorial Region

SAA

START TIME ON TAPE = 1.500000

E. G. STASSINOPOULOS-1981





1

SURFACE OF FINITE ALUMINUM SLAB SHIELDS

2

FINITE ALUMINUM MEDIUM)

Figure 155

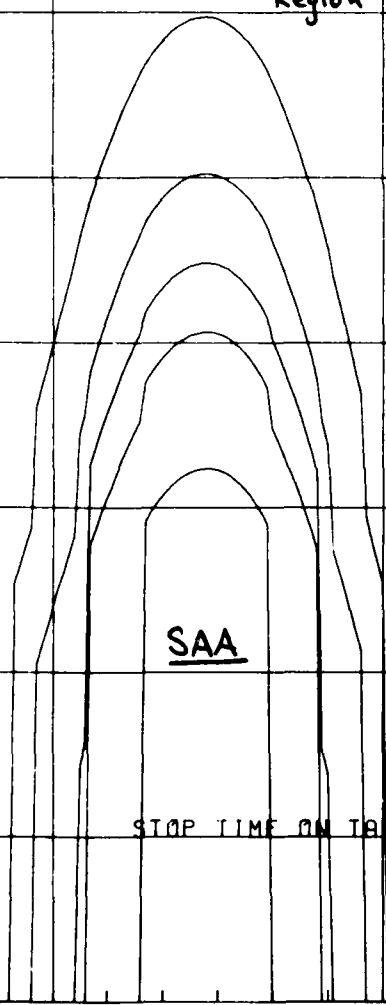
ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Equatorial  
Region



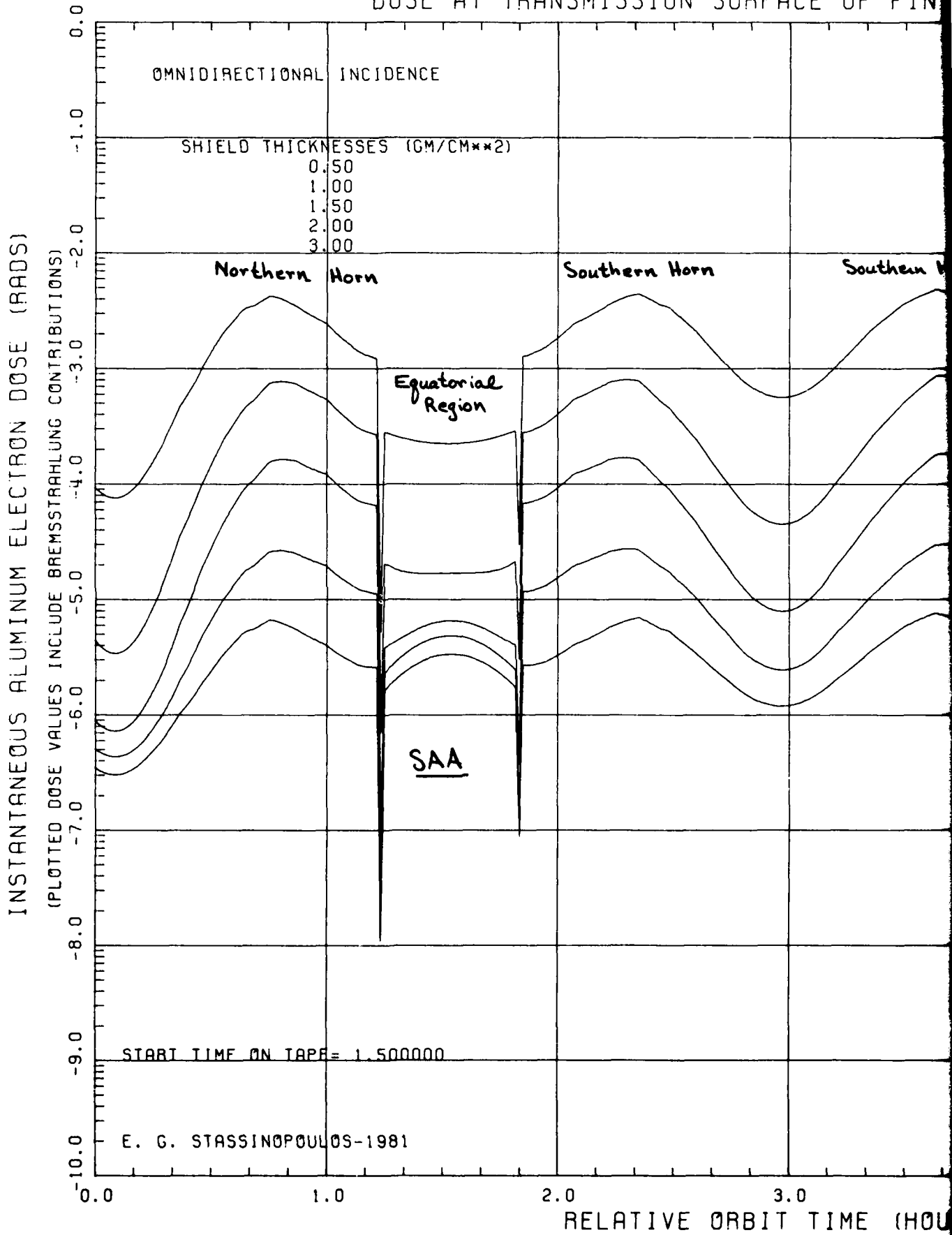
SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

3.0 4.0 5.0 6.0  
ORBIT TIME (HOURS)

DOSE AT TRANSMISSION SURFACE OF FIN



# IN SURFACE OF FINITE ALUMINUM SLAB SHIELDS

Figure 156

ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0

MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Horn

Southern Horn

Equatorial  
Region

Northern  
Horn

SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

3.0

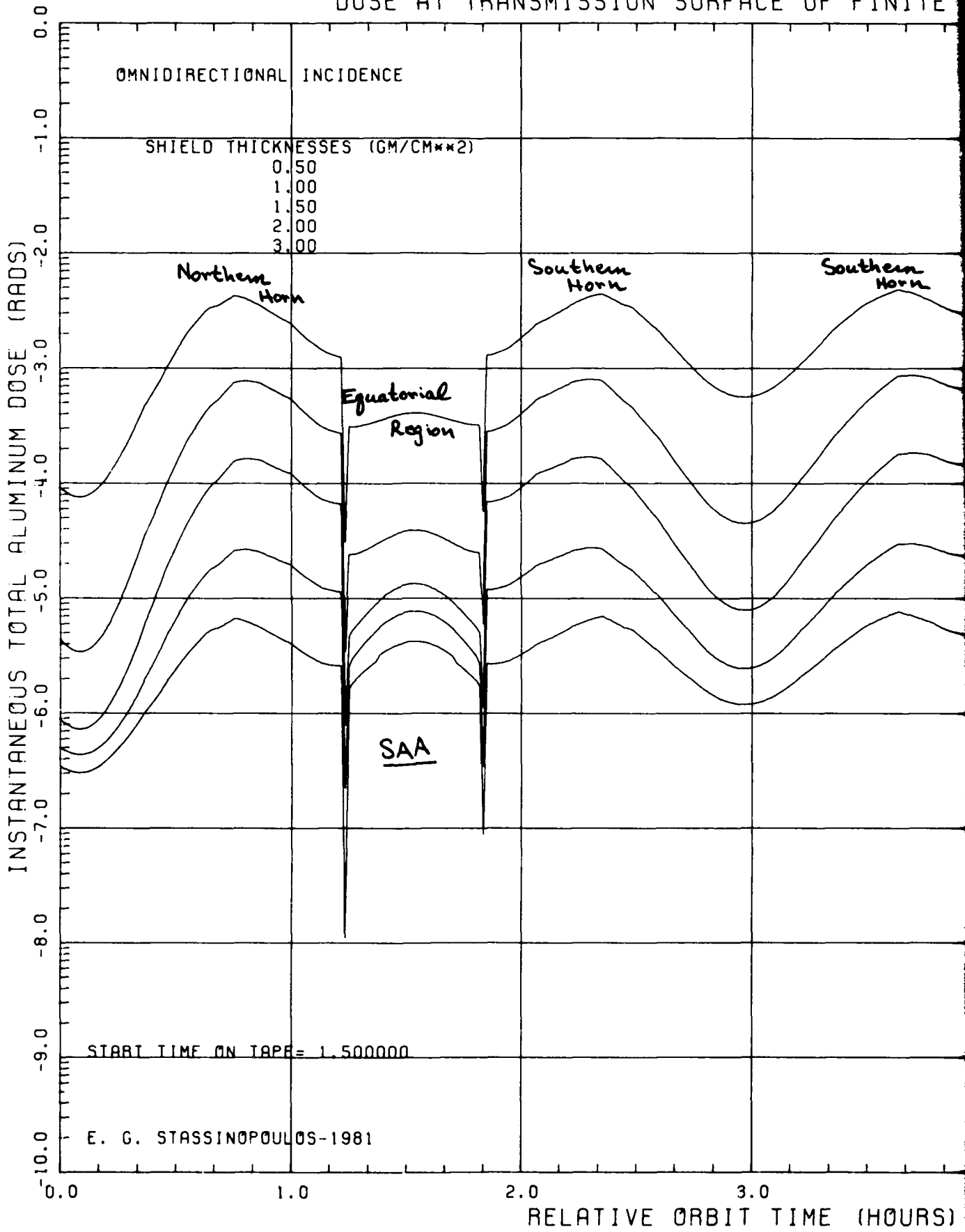
4.0

5.0

6.0

ORBIT TIME (HOURS)

# DOSE AT TRANSMISSION SURFACE OF FINITE



# SURFACE OF FINITE ALUMINUM SLAB SHIELDS

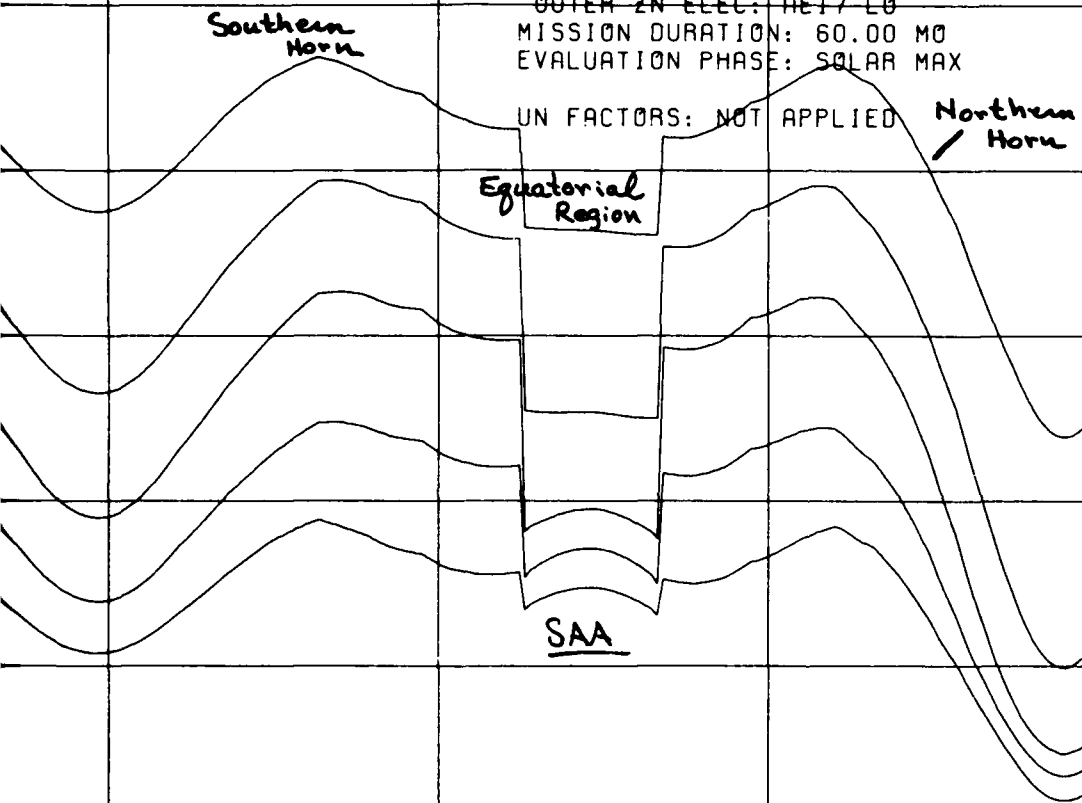
Figure 157

ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED



STOP TIME ON TAPE = 7.483318

NASA-GSFC

3.0 4.0 5.0 6.0

ORBIT TIME (HOURS)

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)

DOSE IN SEMI-INFINITE ALUM

OMNIDIRECTIONAL INCIDENCE

SHIELD THICKNESSES (GM/CM\*\*2)

- 0.50
- 1.00
- 1.50
- 2.00
- 3.00

Northern Horn

Southern Horn

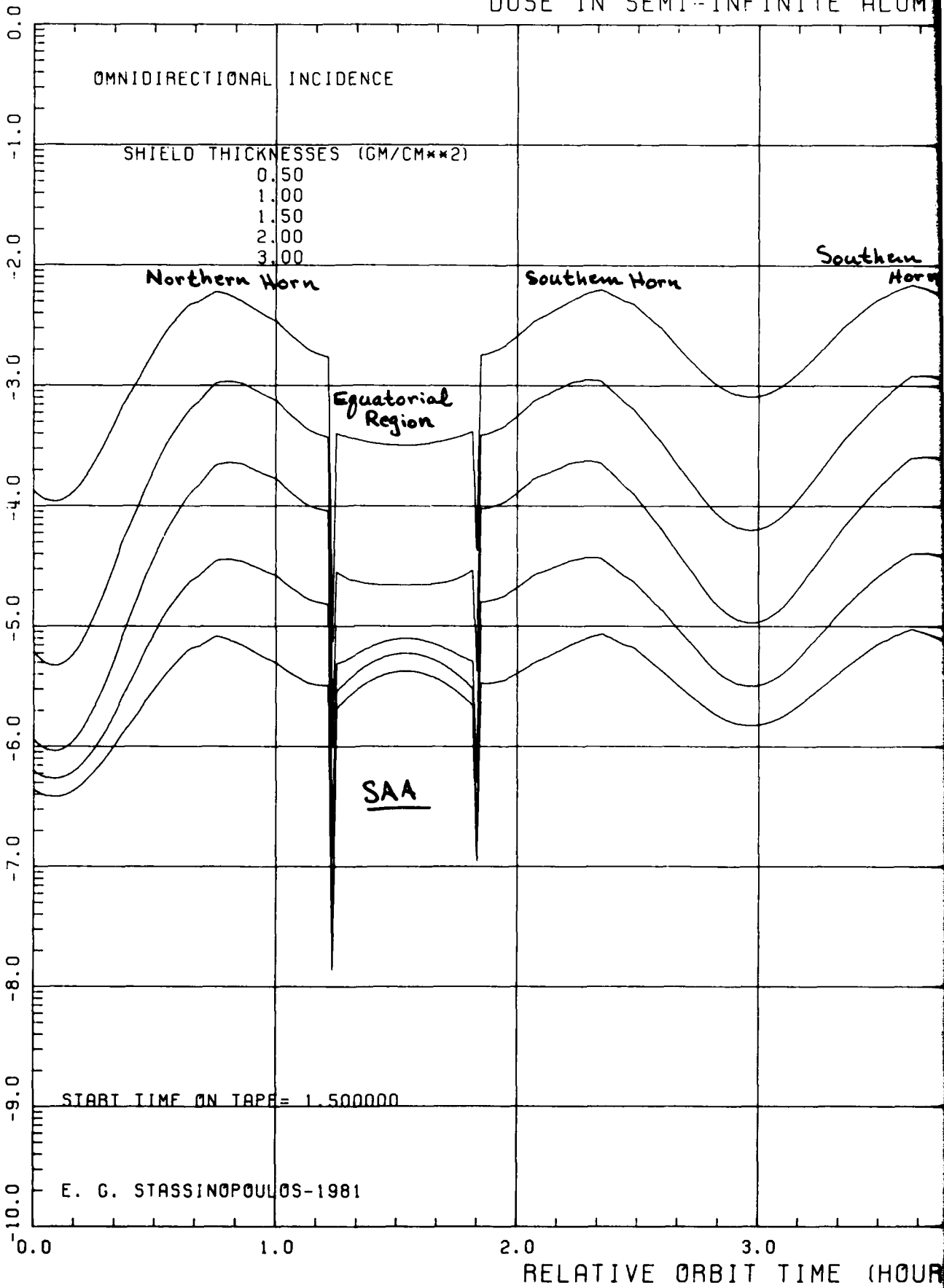
Southern Horn

Equatorial Region

SAA

START TIME ON TAPE = 1.500000

E. G. STASSINOPOULOS-1981



-INFINITE ALUMINUM MEDIUM

Figure 158

ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Southern  
Horn

Northern  
Horn

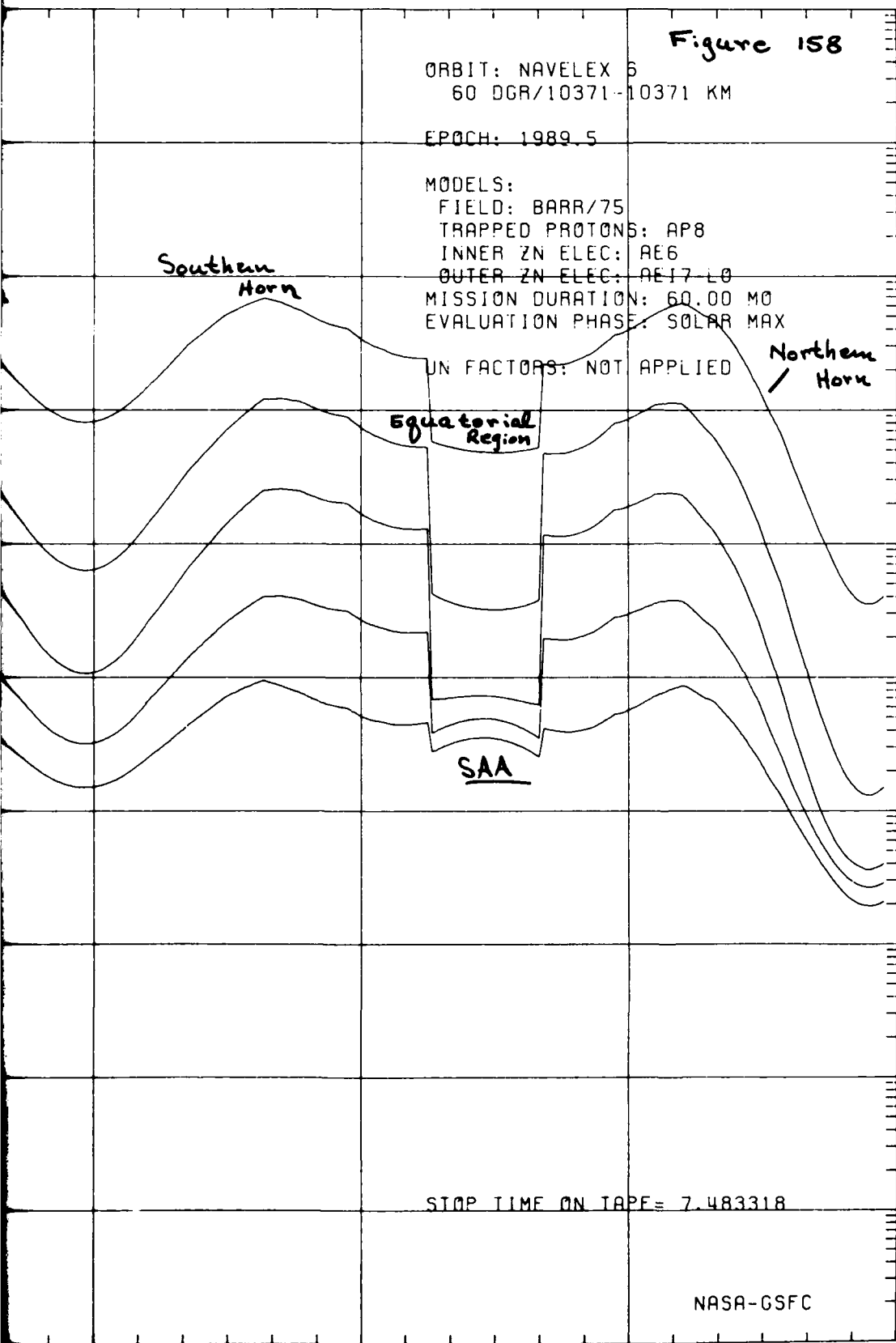
Equatorial  
Region

SAA

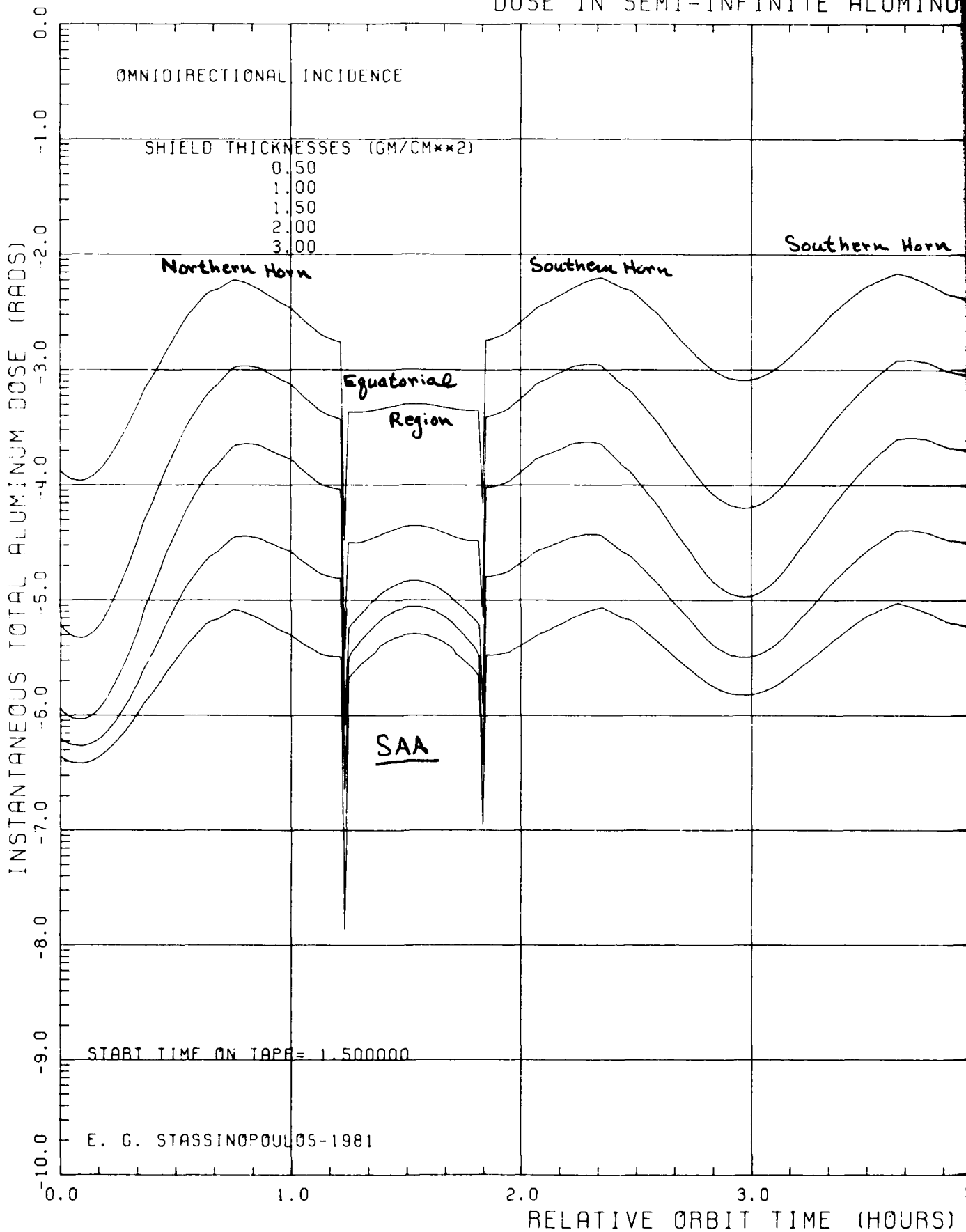
STOP TIME ON TAPE = 7.483318

NASA-GSFC

ORBIT TIME (HOURS)



DOSE IN SEMI-INFINITE ALUMINU





INFINITE ALUMINUM MEDIUM

Figure 159

ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: A65  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Southern Horn

Northern Horn

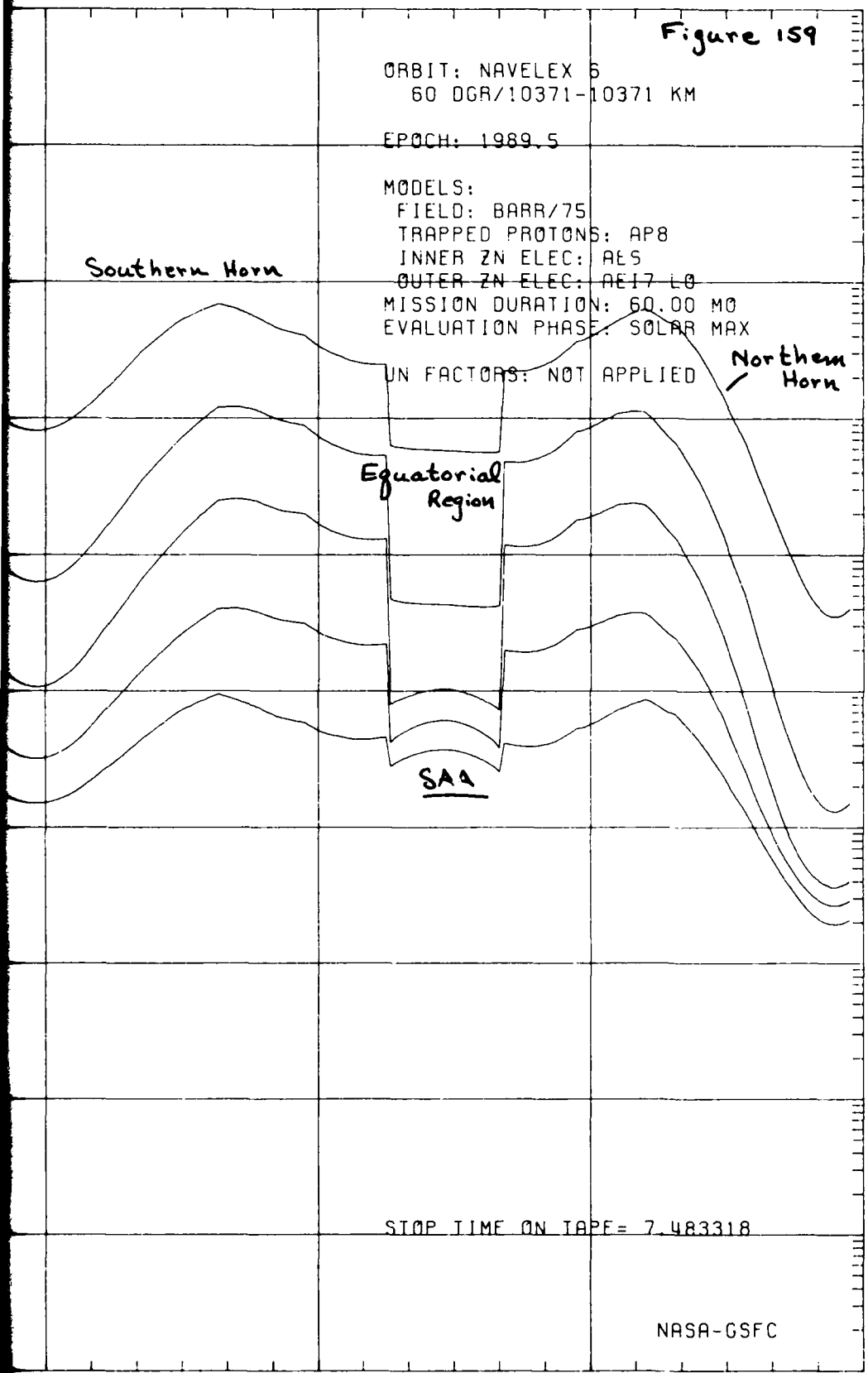
Equatorial Region

SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

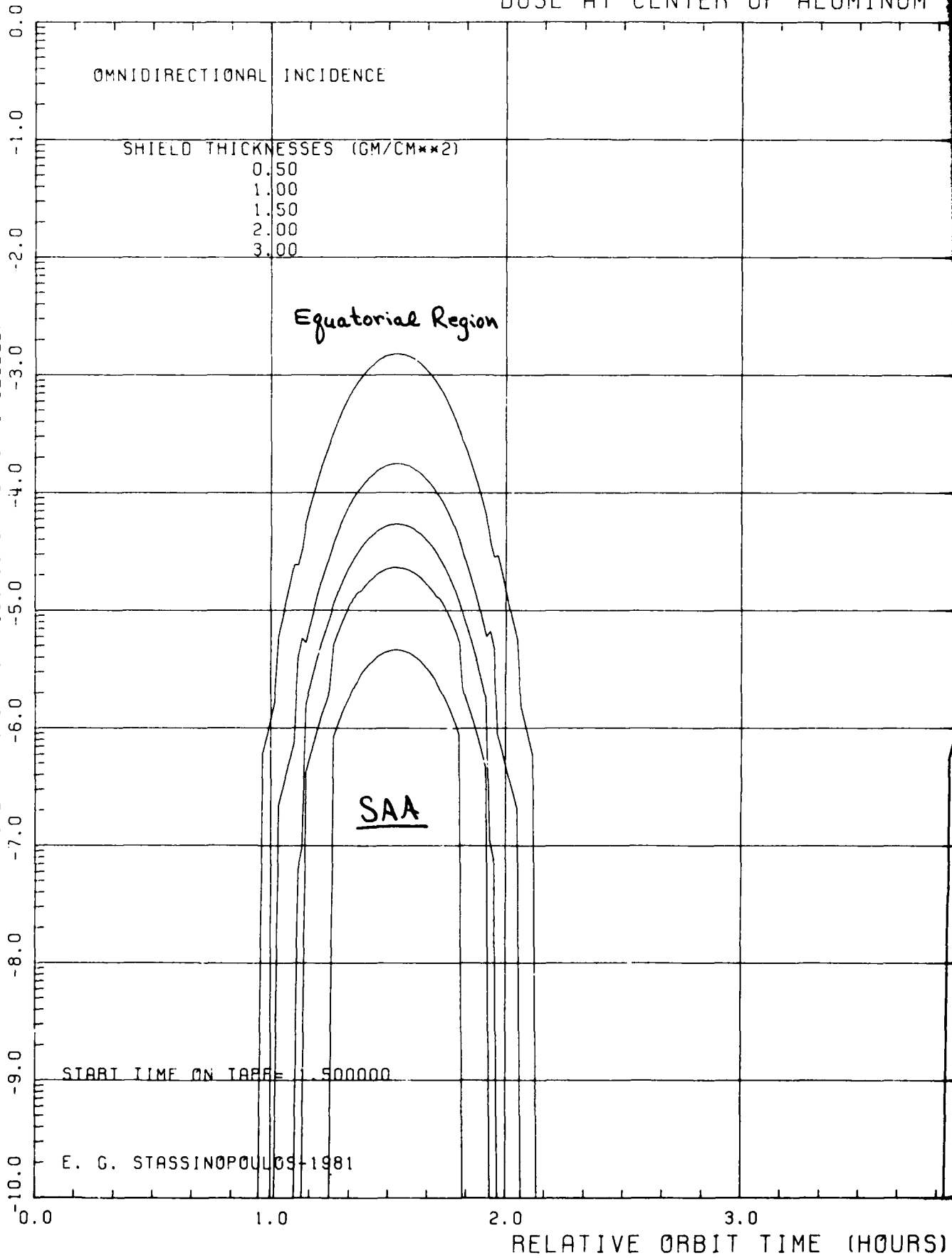
3.0 4.0 5.0 6.0  
IT TIME (HOURS)



DOSE AT CENTER OF ALUMINUM

INSTANTANEOUS ALUMINUM PROTON DOSE (RADS)

(SOLAR PROTON CONTRIBUTIONS ARE NOT INCLUDED)



TER OF ALUMINUM SPHERES

Figure 160

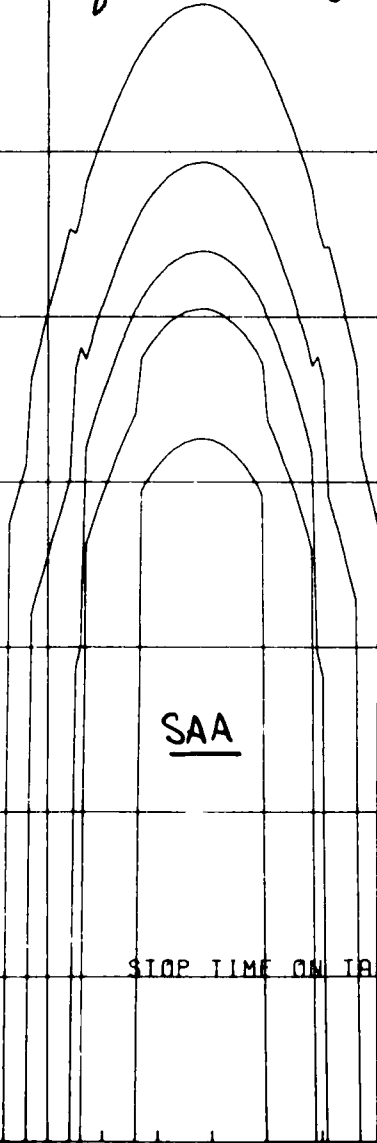
ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17-L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

*Equatorial Region*



SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

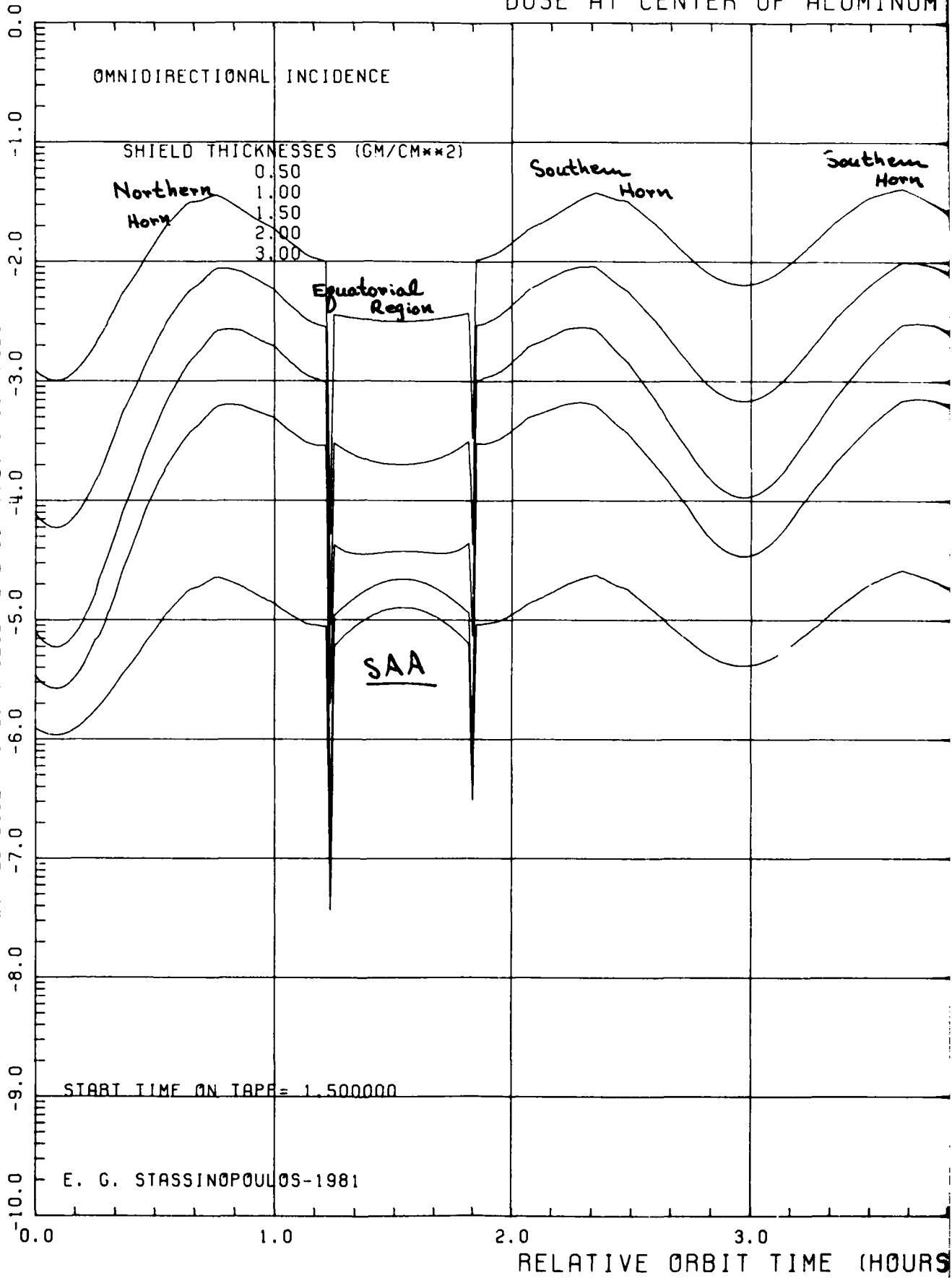
3.0 4.0 5.0 6.0

BIT TIME (HOURS)

DOSE AT CENTER OF ALUMINUM

INSTANTANEOUS ALUMINUM ELECTRON DOSE (RADS)

(PLOTTED DOSE VALUES INCLUDE BREMSSTRAHLUNG CONTRIBUTIONS)



NUMBER OF ALUMINUM SPHERES

Figure 161

ORBIT: NAVELEX 6  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UNCERTAINTY FACTORS: NOT APPLIED

Southern  
Horn

Northern  
Horn

Equatorial  
Region

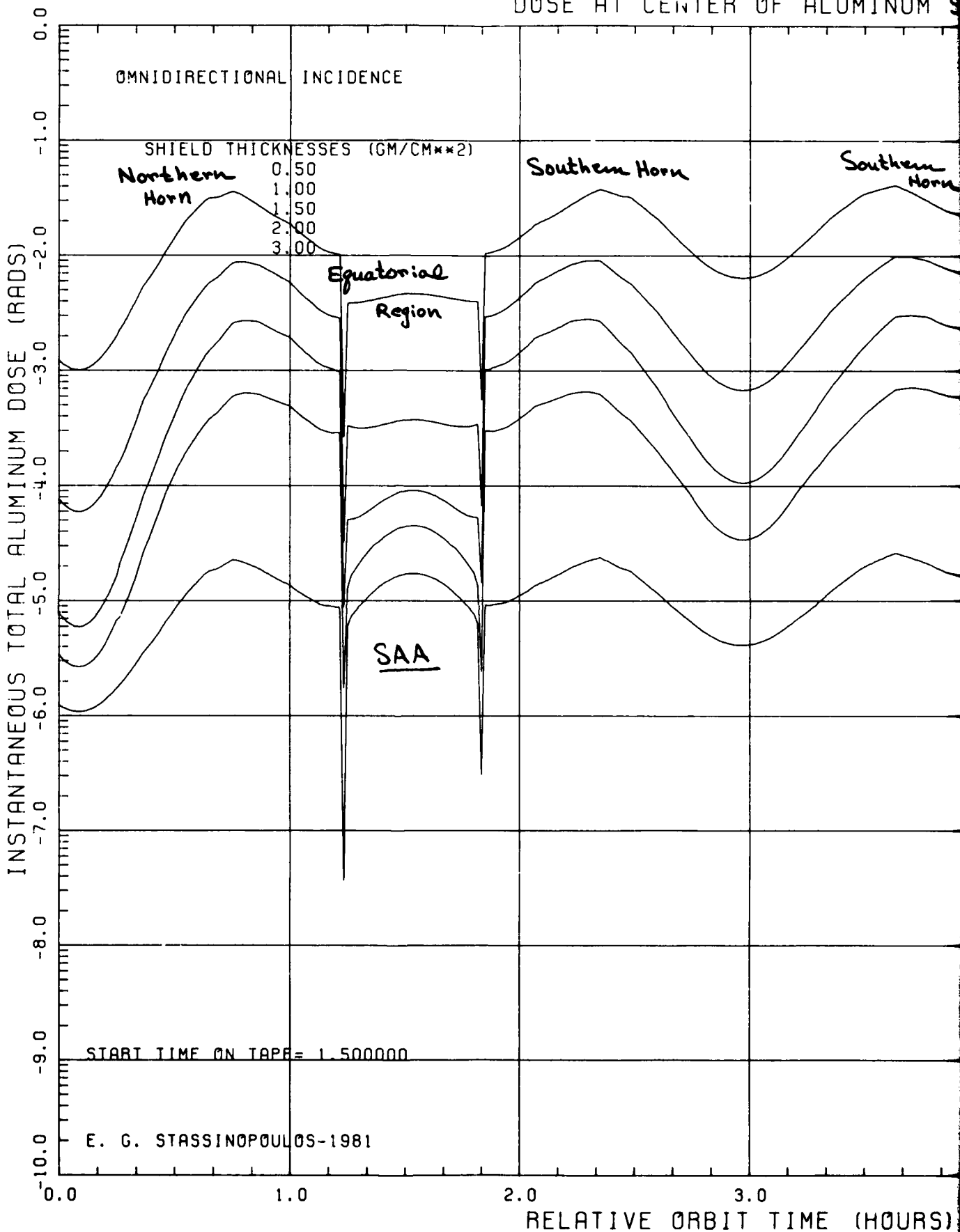
SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

ORBIT TIME (HOURS)

DOSE AT CENTER OF ALUMINUM S



OF ALUMINUM SPHERES

Figure 162

ORBIT: NAVELEX 5  
60 DGR/10371-10371 KM

EPOCH: 1989.5

MODELS:  
FIELD: BARR/75  
TRAPPED PROTONS: AP8  
INNER ZN ELEC: AE6  
OUTER ZN ELEC: AE17 L0  
MISSION DURATION: 60.00 MO  
EVALUATION PHASE: SOLAR MAX

UN FACTORS: NOT APPLIED

Southern  
Horn

Northern  
Horn

Equatorial  
Region

SAA

STOP TIME ON TAPE = 7.483318

NASA-GSFC

3.0 4.0 5.0 6.0  
T TIME (HOURS)

**DATA  
FILM**

**7**