P	AD-A13	6 967	FIL	TILMMISE CONCENSATION OF STEAM ON EXTERNALLY-FINNED 1/1										
	UNCLĂS	SIFIED	CA	w м р	A M PUULE DEC 83 NF569-83-003 N5F-MEH82-03567 F/G 20/13								NL	
			् . १											
					e i s									
					END									- <u> </u>
~	-												-	/



- 1

MICROCOPY RESOLUTION TEST CHART NATIONAL BUREAU OF STANDARDS-1963-A

> . •••

Bead

1. 1. 1.

「「「「おんない」、人名の後の前の一、「あんのあるの」、「ころのあるとう」

SALA ALASSA

ALC:

NPS69-83-003

STATIC STATES

NAVAL POSTGRADUATE SCHOOL Monterey, California



THESIS

FILMWISE CONDENSATION OF STEAM ON EXTERNALLY-FINNED HORIZONTAL TUBES

by

William M. Poole

December 1983

OTIC FILE COPY

Thesis A	Advisor:
----------	----------

P. J. Marto

Approved for public release; distribution unlimited.

Prepared for: National Science Foundation Division of Engineering Washington, DC 20550

JAN 1 9 1984



KEPUKI UULUMENIA	READ INSTRUCTIONS BEFORE COMPLETING FORM	
REPORT NUMBER	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
NPS69-83-003	AD A136967	
TITLE (and Substitio)		S. TYPE OF REPORT & PERIOD COVERE Master's Thesis:
Filmwise Condensation	of Steam on	December 1983
Externally-Finned Hori	Zontal lubes	6. PERFORMING ORG. REPORT NUMBER
		B. CONTRACT OF GRANT NUMBER(#)
William M. Doolo		
WIIIIam M. Poole		MEA82-03567
	NODRESS	10. PROGRAM ELEMENT. PROJECT, TASK
Naval Postoraduate Sch	001	AREA & WORK UNIT NUMBERS
Monterey, California	93943	
CONTROLLING OFFICE NAME AND ADDR	[55	12. REPORT DATE
Naval Postgraduate Sch	001	December 1983
Monterey, California	93943	13. NUMBER OF PAGES
MONITORING AGENCY NAME & ADDRESS	il different from Controlling Office)	15. SECURITY CLASS. (of this report)
		UNCLASSIFICATION DOWNGRADING
Approved for public re DISTRIBUTION STATEMENT (of the ebetree) elease; distribution t entered in Block 20, 11 different from	on unlimited.
Approved for public re OISTRIBUTION STATEMENT (of the ebetree	elease; distributio	on unlimited.
Approved for public re DISTRIBUTION STATEMENT (of the obstree DISTRIBUTION STATEMENT (of the obstree SUPPLEMENTARY NOTES	elease; distributio	on unlimited. m Report)
Approved for public re DISTRIBUTION STATEMENT (of the ebetween DISTRIBUTION STATEMENT (of the ebetween SUPPLEMENTARY NOTES KEY WORDS (Centinue on reverse elde if nor Filmwise Condensation	elease; distribution t entered in Block 20, if different fro eccary and identify by block number;	on unlimited.
Approved for public re DISTRIBUTION STATEMENT (of the about DISTRIBUTION STATEMENT (of the about SUPPLEMENTARY NOTES KEY WORDS (Continue on reverse elde II nor Filmwise Condensation Externally-Finned Cond	elease; distribution t entered in Bleck 20, if different fro eccary and identify by block number; denser Tubes	on unlimited. m Report)
Approved for public re OISTRIBUTION STATEMENT (of the ebotted DISTRIBUTION STATEMENT (of the ebotted SUPPLEMENTARY NOTES KEY WORDS (Continue on reverse elde II not Filmwise Condensation Externally-Finned Cond Vacuum Condensing Appa	elease; distribution t entered in Block 20, if different fro eccery and identify by block number; denser Tubes aratus	on unlimited.
Approved for public re DISTRIBUTION STATEMENT (of the Report DISTRIBUTION STATEMENT (of the observed SUPPLEMENTARY NOTES KEY WORDS (Continue on revorce elde II nor Filmwise Condensation Externally-Finned Cond Vacuum Condensing Appa Heat Transfer Instrumented Tube	elease; distribution t entered in Bleck 20, if different fro eccary and identify by block number; denser Tubes aratus	on unlimited. m Report)
Approved for public re OISTRIBUTION STATEMENT (of the ebourse DISTRIBUTION STATEMENT (of the ebourse BUPPLEMENTARY NOTES KEY WORDS (Continue on reverse elde II not Filmwise Condensation Externally-Finned Cond Vacuum Condensing Appa Heat Transfer Instrumented Tube ABSTRACT (Continue on reverse elde II neces	elease; distribution t entered in Block 20, if different fro eccary and identify by block number; denser Tubes aratus	on unlimited.
Approved for public re OISTRIBUTION STATEMENT (of the ebourse DISTRIBUTION STATEMENT (of the ebourse SUPPLEMENTARY NOTES KEY WORDS (Continue on reverse elde II not Filmwise Condensation Externally-Finned Cond Vacuum Condensing Appa Heat Transfer Instrumented Tube ADSTRACT (Continue on reverse elde II not The film-condensation and six externally-fin 1 mm thick, and pitche 10.0 mm, were experime	elease; distribution t entered in Block 20, if different from eccary and identify by block number; denser Tubes aratus becary and identify by block number; tion characteristi nned tubes having es of 1.5, 2.0, 2. entally tested.	cs of a smooth tube fins 1 mm high and 5, 3.0, 5.0, and
Approved for public re DISTRIBUTION STATEMENT (of the ebourse DISTRIBUTION STATEMENT (of the ebourse SUPPLEMENTARY NOTES KEY WORDS (Centimus on reverse elde if nee Filmwise Condensation Externally-Finned Cond Vacuum Condensing Appa Heat Transfer Instrumented Tube ABSTRACT (Centimus on reverse elde if need The film-condensation and six externally-find 1 mm thick, and pitche 10.0 mm, were experiment A smooth copper to an outside diameter of	elease; distribution t entered in Block 20, 11 different from eccary and identify by block number; denser Tubes aratus becary and identify by block number; tion characteristi nned tubes having es of 1.5, 2.0, 2. entally tested. ube with an active f 19.05 mm, and an	cs of a smooth tube fins 1 mm high and 5, 3.0, 5.0, and length of 133.5 mm, inside diameter of

Unclassified

SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

12.7 mm was first tested to correlate the inside heattransfer coefficient using the Sieder-Tate equation. The leading coefficient for this equation was found to be 0.034 ± 0.001 , and was used to derive the external condensing coefficient for all of the tubes by subtracting the inside and wall resistances from the measured overall resistance. The condensing coefficient was measured, both at atmospheric pressure and vacuum (84 mm Hg), with the heat flux as a variable.

Condensation data taken for the smooth tube were compared with data in the literature to check the reliability of the apparatus and the data-reduction procedures. The data for the finned tubes showed an optimum pitch of 2.5 mm.

Acces	ssion For		
NTIS	GRA&I	M	-
DTIC	TAB	- f i	
Unanı	nounced	Ē	
Just	ification		
		,,	- (*)
By			_
Dist	ibution/		$1 \sqrt{3}$
Avai	lability C	odes	
	Avail and	/or	-
Dist	Special		
A-	1 1		

5-N 0102- LF- 014- 6601

Approved for public release: distribution unlimited.

Filmwise Condensation of Steam on Externally-finned Horizontal Tubes

by

William M. Poole Lieutenant, United States Navy B.S., Marine Engineering U.S. Naval Academy, 1978

Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN MECHANICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL December 1983

Willia n. Pook

Approved by: Thesis Advisor

Author:

Second Reader

Chairman, Department of Mechanical Engineering

Dean of Science and Engineering

ABSTRACT

The film-condensation characteristics of a smooth tube and six externally-finned tubes having fins 1 mm high and 1 mm thick, and pitches of 1.5, 2.0, 2.5, 3.0, 5.0, and 10.0 mm, were experimentally tested.

A smooth copper tube with an active length of 133.5 mm, an outside diameter of 19.05 mm, and an inside diameter of 12.7 mm was first tested to correlate the inside heattransfer coefficient using the Sieder-Tate equation. The leading coefficient for this equation was found to be 0.034 (±) 0.001, and was used to derive the external condensing coefficient for all of the tubes by subtracting the inside and wall resistances from the measured overall resistance. The condensing coefficient was measured, both at atmospheric pressure and vacuum (84 mm Hg), with the heat flux as a variable.

Condensation data taken for the smooth tube were compared with data in the literature to check the reliability of the apparatus and the data-reduction procedures. The data for the finned tubes showed an optimum pitch of 2.5

878 -

TABLE OF CONTENTS

I.	INTE	ODU	CI	! I 0	N	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	11
	λ.	BAC	K (GRC	UN	ĪD	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	11
	Β.	OBJ	EC	TI	V E	2	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	13
II.	DESC	RIP	T J			F	ÀF	PP		ATO	S		•			•			•	•				•	14
	λ.	575	T F	5 M	01	E	RA L	- Ru	 I		_	_		_	_				_	_		_	_		14
	B	CAC	ተት ጥፑ	7 M	TX			 I M T	, NI 4	• • • •	ידר	•	•	•	•	•	•	•	•	•	•	•	•	•	16
	с.	D10	* *	30 30				· TO	. NT				•	•	•	•	•	•	•	•	•	•	•	•	19
	C.	CTC	а п т	AC M	У И	1 1 2				•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	10
	U •	1	1 E T	201 201	1.	נעי					1.2	•	•	•	•	•	•	•	•	•	•	•	•	•	10
		1.	2	201	.18		•	•	•	•	•	•	•	•	•	٠	•	•	•	•	•	•	•	•	10
		2.	C				58.T	:e	53	грі	.ng		•	•	•	٠	•	٠	•	•	٠	•	•	•	19
		3.	V	en	t	٧a	114	e	•	٠	٠	•	٠	•	•	٠	٠	•	٠	•	•	٠	•	٠	19
		4.	Ľ	lan	0	let	tei	: L	.i1	le	•	•	•	•	•	٠	٠	٠	•	٠	٠	•	٠	•	19
		5.	P	?Ie	ss	5 U I	:e	TI	:a 1	ısd	uC	91	•	•	•	٠	•	•	•	•	٠	•	•	•	20
		6.	F	lei	ie	f	۷a	17	9	٠	•	•	•	•	•	•	•	•	٠	•	•	•	•	•	20
		7.	S	ste	a	n- t	:e I	ı pe	IS	atu	re	F	, I O	ba	•	•	٠	•	•	٠	•	•	•	•	22
		8.	C	:00	11	ng	j ii	lat	:e 1	c S	iys	te		•	•	•	•	•	•	•	٠	•	•	•	22
		9.	1	l he		o	pil	.e	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	23
	E.	VAC	00	JM	IN	ITI	EG F	II	Y	•	•	•	•	•	•	•	•	•	•		•	•	•	•	23
	F.	TUB	e :	5 T	ES	5 T I	ZD	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	26
		1.	1	[ns	tI	u	en	ite	bd	Tu	be	•	•	•	•	•	•	•	•	•		•	•		26
		2.	S	5 mc	ot	:h	Tu	ıbe	•	•	•	•	•	•	•	•		•		•	•	•	•	•	29
		3.	F	?in	ne	Ъđ	Tu	۱b∈	s			•			•										29
	G.	SYS	t I	e M	OE	P E I	RAT	IC	N	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	29
			-	-																					30
111.	LTTE	I WT2 -	5				138	TI	.01	N	•	•	•	•	•	•	•	•	•	٠	•	•	•	•	32
	A.	THE	D	RC	PW	IIS	SE	PF	O E	SLE		•	•	•	•	٠	•	•	٠	٠	٠	•	•	•	32
	в.	SOL	UT) N	•	•	•	•	•	٠	•	•	•	•	•	•	•	•	•	•	•	•	٠	32
	C.	VAP	ÔF	2 1	EI	.00	TI	Y'	L	EMT	TA	TI	ON	S		-	•					•		•	35

IV.	DATA	RED	UCTI	(0 N	•	•	• •	•	•	•	•	•	•	•	•	•	•	•	•	•	•	36
	A.	PROG	RAM	SI	EDE	R		•	•	•	•	•	•	•	•	•	•	•	•	•	•	36
	в.	PROG	RAM	WI	lso	N .		•	•	•	•	•	•	•	•	•	•	•	•	•	•	36
	c.	PROG	RAM	DR	P	• •		•	•	٠	•	•	•	•	•	•	•	•	•	•	•	37
۷.	RESU	LTS	A ND	DI	scu	ssi		i .	•	•	•	•	•	•	•	•	•	•	•	•	•	38
	λ.	INSI	DE H	IE A	тт	RAI	I S E	ER	CO	EF	FI	CI	EN	Т	•	•	•	•	•	•	•	38
		1.	Inst	ru	nen	teć	i 1	lube	R	es	ul	τs		•	•	•	•	•	•	•	•	38
		2.	Smoo	oth	Tu	be	Re	sul	ts		•	•	•	•	•	•	•	•	•	•	•	38
		3.	Disc	re	pa n	ci	e s	•	•	•	•	•	•	•	•	٠	•	٠	•	•	•	41
	Β.	FINN	ED 1	UB	ËS	• •	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	45
VI.	CONC	LUSI	ONS	A N	DR	ECC) MR	IENC	AT	'I)	n s		•	•	•	•	•	•	•	•	•	51
	λ.	CONC	LUSI	ON	s	• •		•	•	•	•	•	•	•	•	•	•	•	•	•	•	51
	в.	RECO	MMEN	I D A	TIO	NS	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	51
APPENDI	X A:	TH	ERMO)C O	UPL	EC	C AI	, I BR	AT	IJ	N	•	•	•	•	•	•	•	•	•	•	52
APPENDI	IX B:	SA	MPLE	C C	ALC	ULI	A TI	ONS		•	•	•	•	•	•	•	•	•	•	•	•	55
APPENDI	X C:	51	EAM-	-CL	EAN	INC	3 E	ROC	ED	UR	E	•	•	•	•	•	•	•	•	•	•	58
APPENDI	IX D:	SY	STEN	I S'	TAR	T-1	JP	A ND	s	HU	T -	DO	WN	E	PRO	DCE	DU	RE	s	•	•	59
APPENDI	X E:	co	M PU I	ER	PR	OGI	RAM	I LI	ST	IN	GS		•	•	•	•	•	•	•	•	•	60
LIST OF	? REF	EREN	CES	•	•	• •	• •	•	•	•	•	•	•	•	•	•	•	•	•	•	•	82
INITIAI	. DIS	TRIB	UTIC)N :	LIS	т.		•	•		•	•	•	•	•	•	•	•	•	•	•	84

LIST OF FIGURES

2.1	Schematic of the test apparatus 1	5
2.2	Line diagram of the power supply 1	7
2.3	Calibration of the pressure transducer 2	1
2.4	Schematic of the instrumented tube	
	construction	7
2.5	Photograph of the instrumented tube 2	8
2.6	Photograph of the finned tubes tested 3	0
3.1	Cross-sectional view of the tube and holders 3	4
5.1	Inside Nusselt number plot for the	
	instrumented tube	9
5.2	Wilson plot for the smooth tube 4	0
5.3	Fujii correlation [Ref. 10] for smooth-tube	
	data	2
5.4	Relative heat fluxes for atmospheric and	
	vacuum runs	3
5.5	Comparison of finned tubes with smooth-tube	
	performance	6
5.6	Outside condensing coefficients corrected	
	for area	7
5.7	Variation of condensing coefficient with fin	
	pitch	8
5.8	Sketch of the effect of finning on	
	condensation	0
λ.1	Calibration curve for the thermocouples 5	4

NOMENCLATURE

с <mark>р</mark>	Specific heat, kJ/kg-K
Di	Inner diameter, m
Dø	Outer diameter, m
P	Parameter, gd_h _{ig} /u²kaT
g	Acceleration due to gravity, m/s^2
GrA	Grashof number of air, $g(3(T_w - T_w) L^3/\gamma^2)$
h	Heat-transfer coefficient, W/m²-K
h fg	Latent heat of vaporization, kJ/kg
k	Thermal conductivity, W/m-K
k _A	Thermal conductivity of air, W/m-K
k _G	Thermal conductivity of glass, W/m-K
k_	Thermal conductivity of liquid, W/m-K
L	Length, M
L _S	Length of the boiler, m
Nu	Nusselt number, hdo/k
P	Pressure, MPa
Pr	Prandtl number, µc _r /k
Q	Heat transfer rate, W
g	Heat flux, W/m²
R	Thermal resistance, K/W
Rw	Wall thermal resistance $(D_o \ln (D_o/D_i)/2k_w)$, K/W

r _{i,ð}	Inner radius of the boiler, m
I _{0,8}	Outer radius of the boiler, m
Ra	Rayleigh number, GrPr
Re	Reynolds number, $u_0 D_0 / \nu_0$
Re	Two-phase Reynolds number, $u_{ab}D_{o}/r_{b}$
T	Temperature, [°] C
T _f	Film temperature, °C
TUB	Inner wall temperature of the boiler, $^{\circ}$ C
T _{c,8}	Outer wall temperature of the boiler, $^{\circ}$ C
T 🕳	Ambient temperature, °C
Ŧ	Average temperature, °C
u	Velocity, m/s
U 🕳	Vapor velocity, m/s
Ŭ o	Overall heat-transfer coefficient, W/m^2-K
X	Parameter, $D \neq \frac{1}{3}/2k_{\perp}$
Y	Parameter, $\gamma^{1/3}[1/0_o - R_w]$
Z	Sieder-Tate parameter, Re ^{0.8} Pr ^{1/3} (مراس) 0.14
Ģ	Expansion coefficient of air, K^{-1}
مر	Dynamic viscosity, N-s/m²
*	Kinematic viscosity, m²/s

ACKNOWLEDGEMENT

The author wishes to express his sincere appreciation to his advisor, Professor P. J. Marto, for his advice and timely guidance, without which the author would still be searching for the forest.

Sincere appreciation is also extended to Dr. A. S. Wanniarachchi, for his expertise and for his tireless efforts during the course of this research. LEVERAGE AND A DAMA AND RECORDER AND RELEASED AND RECORDER AND A DAMA AND A DAMA

Appreciation is also extended to the laboratory supervisor, Mr. Fom Christian, the lab technician, Mr. Ron Longuiera, and to the workers in the Mechanical Engineering Department Machine Shop for their efforts.

And thanks to Dcnna for her patience and understanding throughout the course of this effort.

I. INTRODUCTION

A. BACKGROUND

As combat systems aboard naval surface ships continue to grow in importance, size, and weight, the feasibility of their installation aboard lightweight vessels can only be realized by reducing other major shipboard weight requirements. A significant achievement in reducing main propulsion weight was reached with the advent of gas-turbine propulsion plants, of which the majority of our new combatants will be powered. However, for the remaining capital ships and submarinas to be constructed using either conventional or nuclear power plant weight reduction can only be steam systems, realized by increasing the effectiveness of the individual plant components and thereby reducing their size. Plans call for future installation of Rankine-zycle waste-h`at recovery systems even on those ships with gas-turbine propulsion, so significant weight reduction could also be realized there through the use of more effective systems.

Due to the nature of the condensing process, the greatest thermal resistance to condenser-tube heat transfer occurs on the tube side, and a thorough review of tube-side enhancement is presented by Bergles and Jensen [Ref. 1]. Internal enhancement of the tube without regard to the shell-side resistance problem, however, would be a wasted effort if improvements were made to the point where the external (shell-side) resistance became the controlling factor in the transfer of heat. Recent investigations into the shell-side condensing process and techniques to improve this process are thoroughly reviewed by Marto [Ref. 2]. Outside enhancement techniques include the use of low

integral fins, roped tubes, fluted tubes, and applied coatings to promote dropwise condensation.

ander miden Seinen de Seine seine Seine sin in der seine Seine seine seine seine seine seine seine seine seine

Ongoing research is underway at the Naval Postgraduate School to further study the shell-side enhancement of marine condensers. An endurance apparatus is in operation to examine the dropwise-promoting effectiveness of various metallic and polymer coatings over prolonged periods of time. Reilly [Ref. 3] conducted a study of improvements using various spirally fluted tubes under filmwise condensation, and the effects of tube bundle inundation have also been studied by Kanakis [Ref. 4].

SANAN SANANAN ANANANAN SANANAN ANANANAN ANANANA

Southers accord accorded to statistic accord

A test apparatus has been constructed by Krohn [Ref. 5] to systematically study steam condensation on a single horizontal tube. The apparatus is modeled after a similar design constructed by Rose [Ref. 6], but can also operate under the high-vacuum conditions found in marine condensers and without the presence of noncondensable gases. This test apparatus was instrumented and tested by Graber [Ref. 7], who determined the Sieder-Tate coefficient for the test tube length to be 0.029.

In experiments using low integral fins at Queen Mary College of London, Rose, et. al., [Ref. 6] obtained an optimum pitch of 2.0 mm (0.08 in.) for a given fin size of 1 mm (0.04 in) high and 0.5 mm (0.02 in) wide for tests conducted at one atmosphere. With this pitch, he reported vapor-side heat-transfer enhancements of 400% and 300% for heat fluxes of 0.3 MW/m^2 (9.51x10⁺ Btu/hr-ft²) and 0.8 MW/m^2 (2.54x10⁵ Btu/hr-ft²), respectively, and for a vapor velocity of 0.7 m/s (2.3 ft/sec). Further testing is needed to compare these results with similar fin geometries under vacuum conditions, while varying fin dimensions (height and width) in addition to the fin pitch.

B. OBJECTIVE

(1935) MANANA MARAMAN NAMADAN MANANAN AMATANA

The main objectives of this research effort were, therefore, to:

1) ensure a vacuum-tight apparatus so that data could be taken at both atmospheric and vacuum conditions with no detrimental effects due to the presence of noncondensable gases;

2) take data for an instrumented smooth tube to verify the Sieder-Fate coefficient obtained by Graber;

3) take data for a smooth tube to check the reliability of the apparatus and the data-reduction procedures used; and

4) take data for six externally-enhanced tubes of various fin pitches to obtain the relative optimum pitch for a fixed fin geometry.

II. DESCRIPTION OF APPARATUS

A. SYSTEM OVERVIEW

The apparatus used for this research was essentially the same used in references 5 and 7, with several noted modifications. A schematic sketch of the system is shown in Figure 2.1. Steam was generated in a 304.8-mm (12-in) diameter Pyrex glass boiler by ten 4000-watt, 480-wolt Watlow immersion heaters. Passing through a 304.8-mm (12-in) to 152.4-mm (6-in) reducing section, the steam travelled upward through a Pyrex section 2.44 m (8.0 ft) in length, around a 180-degree bend, and back down a straightening section 1.52 m (5.0 ft) in length before entering the stainless-steel test section. The condenser tube to be tested was mounted horizontally in the test section behind a viewport to permit visual observation of the condensing process.

Steam that did not condense on the test tube passed into a stainless steel auxiliary condenser, and all condensate was returned via gravity to the boiler. The auxiliary condenser was constructed of two 9.5-mm (3/8-in) watercooled copper lines helically coiled to a height of 457 mm (18 in).

Cooling water for the test tube was provided by two centrifugal pumps connected in series. The water could be throttled from zero flow to 0.69 1/s (11 gpm). The maximum water velocity which could be obtained through the tube was 5.48 m/s (18 ft/sec). A continuous supply of tap water was used for cooling the auxiliary condenser. Throttling the flow of tap water through the condenser was the means used to vary the internal pressure of the test apparatus. The water flow through both the test tube and the auxiliary



ł

552335291 - 353453491 - 252322331 - 353452531



condenser was regulated by 19.1 mm (3/4 in) diameter needle valves and measured by rotameters with full-scale ranges of 0.69 l/s (11 gpm).

An air ejector provided for removal of noncondensable gases from the auxiliary condenser through a 12.7-mm (1/2-in) line. The source for the air ejector was 1.1-MPa (160-psig) house-air supply.

B. SYSTEM INSTRUMENTATION

and an and an and an and an and an

122222

The input voltage through the heaters was varied through a panel-mounted potentiometer. 440-VAC line voltage was reduced by a factor of 100 when fed into a differential input precession voltage attenuator. The stepped-down voltage passed through a True-Root-Mean-Square converter stage on which the integrated period was reduced to about 1 The output of the TRMS converter was then ms. buffed and compared to a reference voltage from the potentiometer. The comparator output was fed to the control input of a Halmar silicon-controlled rectifier power supply which applied the actual voltage to the heaters. The TRMS converter output was also paralleled to a filter and then input to the data acquisition system. This input was proportional to the power supply output. A diagram of the system is shown in Figure 2.2.

The internal pressure of the system was measured manually by a U-tube, mercury-in-glass manometer graduated in millimeters. Unavoidably, steam could condense in the manometer. Therefore, the varying height of the water column in the mancmeter needed to be accounted for when measuring the system pressure.

Temperatures throughout the system were measured by copper-constantan thermocouples: six for the wall of a specially-constructed test tube, two for the steam, and one

Line diagram of the power supply. Figure 2.2

जन्म, जन्म, मेल, जन्म, जन



.

each for the cooling-water inlet, condensate return, and ambient. The calibration procedure for these thermoccuples is described in Appendix A. The temperature rise through the test tube was measured by a Hewlett-Packard (HP) 2804A quartz thermometer.

All temperature measurements were fed directly into the data-acquisition system as described below.

C. DATA ACQUISITION

1. S. 1. S.

22444444 (4444444) (2200000) (2200000) (220000)

An HP 3497A Data Acquisition/Control Unit was used to monitor system temperatures. This was interfaced with an HP 9826A computer which served as a controlling unit through an interactive data-reduction program and user keyboard prompts. Raw data gathered by the data-acquisition system were stored on computer disks for later reduction and evaluation.

D. SYSTEM MODIFICATIONS

1. <u>Boiler</u>

The fiberglass insulation was removed from the boiler to allow the operator to more easily monitor the water level. Although a closed-system design was used, it was still possible for steam to escape via the air ejector or through the relief valve [Fig. 2.1]. Calculations showed that the additional heat loss due to the removal of the insulation was minimal (App. B], and the author felt this loss was much more acceptable than risking damage to the immersion heaters through a low-water casualty in the boiler.

2. <u>Condensate Piping</u>

As originally designed, draining the system required breaking down the condensate piping. While this was not a daily occurrence, the procedure was inconvenient and there existed also the possibility of losing the vacuum integrity of the system each time it was done.

To avoid these problems, the existing fill-line valves were rearranged as shown in Figure 2.1. This arrangement also added two additional features to the system:

1) the fill/drain valve could be opened during operation to drain any heavy particulate matter from the system similar to the "bottom blow" procedure used on Naval boilers; and

2) after extended periods of inactivity while opened to the atmosphere, the entire system could be given a thorough steam-cleaning by following the procedures outlined in Appendix C.

3. Vent Valve

The modification of the condensate return piping necessitated the addition of a vent value for use when filling or draining the system. A 4.3-mm (0.17-in) needle walve was installed on the 101.6-nm (4-in) flange of the test section. This value would also serve as the tap for the proposed sampling of noncondensable gas concentrations in the system.

4. <u>Manometer Line</u>

The original system design used a 6.4-mm (1/4-in) stainless steel tube angled down to the mercury manometer. During this thesis, the manometer was raised to eye level to facilitate easier and quicker reading. Replacement of the stainless-steel line by a 12.7-mm (1/2-in) copper tube reduced the possibility of error caused by water slugs building up in the smaller diameter tube. The more workable copper was chosen over stainless steel to reduce the stiffness of the connecting line. This was necessary to eliminate leakage in this part of the apparatus as explained in Section II.E.

5. Pressure Transducer

(AACA) SUSSIONE STRATED PARADAD SALADADAD

244444491 ASAU0091 SESSESSES (XAAPAAA) 55

As an alternative to the manometer, a Celesco strain-gage pressure transducer was installed on the test section flange next to the vent valve. The calibration line for the transducer is shown in Figure 2.3. The author felt, however, that the reliability of this measurement would not be high enough until a second, more accurate transducer was installed. Once incorporated into the system, though, these transducers would provide automatic input to the data acquisition system, eliminating the requirement to manually enter the manometer reading into the data-reduction program.

6. <u>Relief Valve</u>

As originally designed, a 6895-Pa gage pressure relief valve was installed beneath a 1.0-m (1.0-psig) (39.4-in) length of 12.7-mm (1/2-in) stainless-steel piping. Steam which condensed and became trapped in this piping would open the valve at a water-column height of only 0.70 m (27.6 in). Once opened, a back-pressure of 0.14 MPa gage (20 psig) was required to reseat the valve - something unobtainable even with an absolute vacuum on the inlet side of the valve. To avoid this problem, the valve was raised in the line to a point only 76 mm (3.0 in) below the outlet from the auxiliary condenser section and was replaced by another 6895 Pa gage pressure (1.0-psig) relief valve which reseated at only 0.04-MPa gage (6-psig) back-pressure.



(2002)

ر ۲ د

Figure 2.3 Calibration of the pressure transducer.

7. <u>Steam-temperature Probe</u>

The steam temperature probe was located directly above the test tube and shared the same inlet to the test section as the manometer line. When drawing a high vacuum, this arrangement allowed water which had collected in the manometer to be drawn back into the test section where it would flow down the probe and onto the tube. To prevent this water and any contaminants picked up in the manometer from being deposited on the tube, the probe was bent so that it was offset from the center of the test tube.

8. Cooling Water System

ANY AND SECOND REPORT OF THE PARTY OF

A second centrifugal pump was added in series to the one already installed. This pump boosted the maximum cooling water velocity through the test tube from 3.96 m/s (13 ft/sec) to 5.48 m/s (18 ft/sec).

The 10.55 kW (3 Ton) air-conditioning unit used with the cooling water system would energize at a water temperature of 17 °C (62.6 °F) and secure when the temperature was reduced to 13.4 °C (56.1 °F). Therefore, for a given steam temperature of 50 °C (122 °F) around the tube, the logmean-temperature difference would vary by as much as 11%. But the measured temperature rise of the water through the tube showed very little change. To avoid this transient problem, the air-conditioning unit was not used, and instead fresh tap water was continuously fed to the sump while an equal amount of water was being drained, maintaining a constant sump level. This method provided a constanttemperature supply of cooling water to the inlet side of the test tube.

It should be pointed out that, while the duty cycle of the air-conditioning unit was a function of the thermostat used, a more sensitive thermostat would require the

use of a more expensive cooling device than a commercial air conditioner. An alternative solution would be to install a much larger sump in the system.

9. Thermopile

Since the temperature rise of the cooling water through the tube was a critical measurement in the experiment, a 10-junction thermopile was added to measure this temperature rise in addition to the quartz thermometer. As will be explained in Section V.A.3, however, problems arose with the thermopile, and it could not be accurately used during this thesis.

E. VACUUM INTEGRITY

One of the objectives of this thesis was to ensure a vacuum-tight test apparatus to eliminate the presence of any noncondensable gases and their detrimental effects on the condensing process. A standard of no more than a 5.0-mm mercury (0.10-psi) loss over a 24-hr period was considered to be an acceptable tolerance, but obtaining a leak rate within this tolerance proved to be the most time-consuming effort during the research.

Due to the construction of the apparatus and nature of the experiment, most leak-detection methods could not be used. A sealing substance could not be used without risking contamination of the interior of the apparatus which might prohibit filmwise condensation on the test tube.

Initially the system was pressurized and the standard soap-solution test was used to locate leaks. Once pressurized, a liquid-soap solution was applied to each external joint or fitting where a leak could be present. The higher pressure air inside the apparatus would escape through any leaks and produce bubbles on the applied soap film. However, the maximum pressure which the Pyrex glass members could tolerate was only 0.074 MPa gage (10.7 psig), and for safety reasons the author chose not to pressurize the system to more than 0.034 MFa gage (5.0 psig) - a pressure difference between the atmcsphere and the apparatus of only 0.040 MPa (5.8 psi). The numerous externa' valves and fittings prohibited the use of an evacuated hood to achieve a greater pressure difference.

Ì

A similar test could not be used to locate any vacuum leaks which were not present when the system was under a positive pressure, as the apparatus was not large enough to permit the application and observation of a soap solution on the interior.

In another attempt to locate the leaks, a National Research Corporation (NRC) 101.6-mm (4-in) vacuum pumping system was connected to the apparatus. This pumping system included a Welch model 1376M mechanical pump and a model NHS-4 diffusion pump. An NRC model 521 thermocouple gage was also connected to the test apparatus and the entire apparatus was evacuated to 0.21 torr (4.1x10⁻³ psia). Acetone was sprayed around all flanges, fittings, and joints. A leak around any of these should have produced a rapid rise in the thermocouple reading, but this method also proved ineffecprobably due to the large size of the apparatus tive. resulting in too great a mean free path for the acetone moleculas to travel from the leak to the thermocouple.

The next alternative was to break the system apart into three main sections: the glass boiler and steam piping, the stainless-steel test section and auxiliary condenser, and the condensate return piping. The glass section was blankflanged and evacuated to an absolute pressure of 0.033 torr (6.4x10 - 4 psia). The rate-of-rise measurement for this section showed a loss of only 0.48 mm of mercury (0.01 psi) over a 30-hr period. It was, therefore, concluded that any leak in the assembled apparatus was not from this section. Once removed, the test section and the auxiliary condenser were blank-flanged, pressurized to 0.10 MPa gage (15 psig), and immersed in a large plexiglas tank filled with water. This test easily revealed a number of small leaks about the inlet side of the test tube and also around the plug which connected the condensate return piping to the base of the auxiliary condenser. Replacing an O-ring in the tube fitting and silver brazing the plug into place eliminated these leaks.

The same immersion test revealed small leaks in the joints of the condensate return piping. These leaks were eliminated by replacing all stainless-steel ferrules in the Swagelok fittings with teflon ferrules.

Once reassembled, considerable leakage was still indicated by a substantial overnight rise in the manometer lavel. The author felt confident that this leak was not in the main assembly of the apparatus, but was instead in the manometer assembly itself since it could not be pressurized for testing. As mentioned in section II.C.4, the stainlesssteel line leading to the manometer was at this time replaced by a 12.7-mm (1/2-in) soft copper tube. This tube eliminated the need for two 90-degree elbows and three lengths of stainless steel tubing in the line - an assembly which proved too rigid to allow even the slightest misalignment into the manometer.

Upon completion of the installation of this assembly, the system was evacuated to an absolute pressure of 92.5 mm Hg and over a 24-hr period the mercury level rose to only 94.0 mm. This leak rate was well within the acceptable tolerance.

P. TUBES TESTED

1. Instrumented Tube

An instrumented tube was fabricated from a thickwalled copper tube with an inner diameter of 12.70 mm (1/2 in) and an outer diameter of 19.05 mm (3/4 in). The tube was cut into three sections into which six holes were drilled axially along the walls at equal spacings 60 apart. These passages were fitted with 0.094-mm (3/32-in) OD capillaries [Fig. 2.4] which were silver-soldered into place, and the three sections of tube were then soldered back into one piece. Thermocouples were fitted into the capillary sections to measure an average wall temperature.

By knowing the average wall temperature, the Nusselt number for the inside could be computed. By computing the gradient of the Nusselt number against the Sieder-Tate parameter, the inside coefficient could be obtained as the inverse of the gradient. Figure 2.5 shows a photograph of the instrumented tube with the installed thermocouples.





2?



Ŀ

2

Figure 2.5 Photograph of the instrumented tube.

2. Smooth Tube

A smooth tube with no wall thermocouples was also tested to obtain the inside heat-transfer coefficient through the use of a modified Wilson Plot [Ref. 8]. Determination of the inside heat-transfer coefficient was critical to the experiment, as it was used to obtain the outside condensing coefficient for the smooth tube and all of the enhanced tubes.

3. Finned Tubes

To fulfill the main objective of this thesis, a series of six finned tubes was also tested [Fig. 2.6]. These tubes had the same overall dimensions as those above, but were enhanced with radial fins 1 mm (0.04 in) high and 1 mm (0.04 in) thick. Each tube had a different fin pitch and was tested to determine a relative optimum pitch. Fin pitches tested were 1.5, 2.0, 2.5, 3.0, 5.0, and 10.0 mm.

G. SYSTEM OPERATION

The tube to be tested was cleaned in a warm solution of Sparkleen and then rinsed with tap water, which produced a contaminant-free, wetted surface. The tube was then installed in the test section, care being taken not to touch or contaminate the condensing surface.

The system was brought to operating pressure by following the procedures of Appendix D, and data collection began when steady-state conditions were achieved. Steady-state conditions were determined by observing the steam temperature measured by the respective thermocouples. When their output voltage on the HP 3497A reached a constant value with fluctuations of only one or two microvolts, it was assumed that steady-state conditions existed in the test apparatus.



and a

Figure 2.6 Photograph of the finned tubes tested.

Data sets were taken starting with a test-tube cooling (which corresponded to a water water flow rate of 90% velocity of 4.95 m/s or 16.2 ft/sec), ranging downward in decrements of 10% through a minimum flow of 20% (1.16 m/s, 3.8 ft/sec), and then upward from 25% to 85% in increments of 10%. After adjusting the flow rate, the temperature rise through the tube was monitored by observing the digital output of the quartz thermometer. When this rise became constant, another set of data could be taken. During data runs, a slight rise in pressure accompanied the decrease in the cooling water flow through the tube, a result of the reduced heat flux. Similarly, increasing the flow through the tube caused a slight decrease in the system pressure. This variance could be anticipated, and since data were to be collected at a constant pressure, it was easily compensated for by throttling the flow of cooling water through the auxiliary condenser one or two percentage points on the rotameter.

Something which could not be anticipated, however, were sudden fluctuations in the tapwater pressure to the auxiliary condenser which caused pressure changes of several millimeters of mercury in the system. To avoid this problem, the flow through the auxiliary condenser had to be continuously monitored, unless data were being taken late at night when there was no demand on the laboratory building's water supply. The test tube could be easily monitored through the viewport for confirmation of filmwise conditions. If there was any sizeable change to dropwise condensation on the tube, the data set was discarded and the procedure was repeated.

III. FILMWISE CONDENSATION

A. THE DROPWISE PROBLEM

provide and provide and the second

1222222

It was essential during the course of this thesis to collect data under filmwise conditions. Numerous problems were encountered by Graber [Ref. 7] in avoiding the transition to dropwise condensation during operation, and his proposed solution was a vacuum-tight test apparatus. Since the tube surface would wet completely after installation, contaminants leaking into the system were possibly adhering to the surface and promoting the dropwise condensation.

However, even after obtaining a vacuum-tight apparatus, the author was still unable to maintain good filmwise condensation for more than two hours on a smooth tube. While this was enough time to collect a complete set of data for this tube, filmwise condensation lasted seventees minutes at the most on any of the finned tubes, the average time being less than ten minutes. This was predictable - the corners of the fin/surface interfaces provided a better trap for contaminants and were harder to clean - but unacceptable.

The use of the steam-cleaning method described in Appendix C would thorcughly clean the tube so that complete filmwise condensation was re-established, but dropwise condensation would again become prevalent within minutes.

B. SOLUTION

Having eliminated the possibilities of installing a dirty tube or contamination due to leakage, the only reason for the dropwise problem had to be coming from outgassing of the nylon holders for the test tube. The outgassing rate for nylon was found to be almost two orders of magnitude greater than the rate for Teflon [Ref. 9], Teflon being the same order as stainless steel. By punping down the apparatus while it was hot, nylon molecules were being outgassed into the test section and were immediately being deposited onto the surface of the cooler test tube, resulting in inadvertant "sputtering" of the tube with a nylon coating and subsequent dropwise condensation. To eliminate the nylon interface with the interior of the test section, special stainless steel caps were manufactured to slip over the nylon holders.

Teflon bushings were fitted within the caps to insulate them from the the tube. Teflon had both a low thermal conductivity and a relatively low putgassing rate. Figure 3.1 shows a detailed sketch of the installation.

This configuration appeared to solve the problem of dropwise condensation. Although an actual endurance test was not conducted, the system was in operation intermittantly for over fifteen hours with the smooth tube and over four hours with each finned tube with no breakup of the filmwise condensation.

Once installed, this arrangement also eliminated the need for the tube-cleaning procedure recommended by Graber [Ref. 7], who felt that a strong cleaning solution of sodium hydroxide and ethanol was needed to decontaminate the surface. Only a warm solution of Sparkleen was used throughout the data-collection stage of this thesis.


C. VAPOR VELOCITY LIMITATIONS

To check the outside heat-transfer coefficient data with the Nusselt prediction, and also to obtain an accurate inside coefficient with the modified Wilson Plot, Vapor velocities approaching zero were preferred. Due to the design of the system, however, the pressure drop to the auxiliary condenser required vapor flow past the test tube. This being the case, attempts were made to minimize this flow velocity by cutting down the power to the boiler and throttling back on the cooling water supply to the condenser. As vapor velocity was decreased, however, dropwise condensation again took place on the test tube. Under atmospheric pressure, this occurrence was at vapor velocities of about 0.5 m/s (1.6 ft/sec), and under vacuum operations it occurred at about 0.9 m/s (3.0 ft/sec). Apparently there still existed a sizeable rate of outgassing within the test section, those gases collecting about the test tube and interfering with the filmwise condensation process. This suspicion was confirmed when an increase in the vapor velocity eliminated the formation of drops on the tube.

IV. DATA REDUCTION

The data collected and stored on the computer disks reduced using the programs WILSON, SIEDER, and DRP. The programs were amenable to changes, which allowed the author to analyze and compare results while varying parameters within the programs. Stepwise reduction procedures for each program are listed below and the program listings are found in Appendix E.

A. PROGRAM SIEDER

temperature.
re.
•
cooling water.
e cooling water.
mperature.
ifference.
coefficient.

B. PROGRAM WILSON

1. Assume a value for the Sieder-Tate coefficient.

2. Compute the Reynolds and Prandtl numbers for flow through the tube.

3. Compute the log-mean-temperature difference, heat flux, and overall heat-transfer coefficient for the tube.

4. Assume an outer tube surface temperature.

5. Compute the outside condensing coefficient using properties evaluated at the film temperature.

6. Compute the cuter surface temperature and iterate steps 5 and 6 if not within 1%.

7. Assume a viscosity-correction factor for the Sieder-Tate equation of 1.0.

8. Compute the inside heat-transfer coefficient.

9. Compute the inner surface temperature.

10. Compute the viscosity correction factor and iterate steps 8 through 10 if not within 1%.

11. Compute the Sieder-Tate coefficient and iterate steps 2 through 11 if not within 0.5%.

C. PROGRAM DRP

1.	Compute	the	average cooling water temperature.
2.	Compute	the	cooling water velocity.
3.	Compute	the	mass-flow rate of the cooling water.
4.	Compute	the	heat transferred to the water.
5.	Compute	the	log-mean-temperature difference.
6.	Compute	the	overall heat-transfer coefficient.
7.	Compute	the	wall resistance of the tube.
8.	Compute	the	Reynolds number of the cooling water.
9.	Compute	the	inside heat-transfer coefficient.
0.	Compute	the	condensing heat-transfer coefficient.

V. RESULTS AND DISCUSSION

Numerous data runs were made using the procedure described in Section II.G. Time constraints, however, limited the number of repeat runs that could be made for this thesis. Primary concern focused on establishing a reliable, repeatable Sieder-Tate coefficient. Data were taken for all of the tubes at both atmospheric and vacuum (88mm Eg, 1.7 psia) conditions. Complete filmwise condensation was maintained for all data runs, and the mass concentration of noncondensable gases was held between 0% and -1% during all testing. The negative value was indicative of slight superheat in the system or an inaccurate manometer reading.

A. INSIDE HEAT TRANSFER COEFFICIENT

ういいいいい

APPENDED I REPORTED PROPERTY [LEWISCON

1. Instrumented Tube Results

Figure 5.1 shows the variation of the Nusselt number as a function of the Sieder-Tate parameter for the instrumented tube run at atmospheric pressure (run SDA7). This method yielded a Sieder-Tate coefficient of 0.035 on two (SDA7 and SDA8). separate data runs The same method under vacuum conditions yielded a coefficient of 0.037, which was also repeated (runs SDA5 and SDA6). The temperature distribution around the tube wall was symmetrical about the vertical plane passing downward through the centerline of the tube, and showed as much as a 16 C temperature drop from the top of the tube to the bottom.

2. <u>Smooth Tube Results</u>

Figure 5.2 shows the modified Wilson Plot for smooth-tube data collected at atmospheric pressure (run





Inside Nusselt number plot for the instrumented tube. Figure 5.1





OD12). This method produced a Sieder-Tate coefficient of 0.033 with an intercept of 1.16. A similar plot for vacuum conditions (run OD10) produced a coefficient of 0.036 and an intercept of 1.27. The Nusselt theory predicts an intercept of 1.53 but, as was noted in Section III.C., a finite vapor velocity past the tube was unavoidable in this research.

Figure 5.3 shows a plot of the Fujii [Ref. 10] correlation for the smooth-tube runs. This experimental correlation is to correct the Nusselt number for the effects of vapor shear on the test tube and is a plot of the equation:

$NuRe^{-1/2} = 0.96F^{1/5}$

The results obtained were slightly higher than those predicted using this correlation.

3. <u>Discrepancies</u>

The original Sieder-Tate equation [Ref. 11] for fully developed turbulent flow in a tube with an L/D ratio greater than 60 has a leading coefficient of 0.027, so a higher value for the shorter tube (L/D = 18) used for this research stands to reason. The data for the tubes appears at first inconsistent, with results of 0.033, 0.035, 0.036. and 0.037 - showing up to a 6% scatter from the mean. Both tubes, however, showed a larger coefficient when the test section pressure was reduced, which reduced the heat flux as Dropping the pressure from atmospheric to a high well. vacuum (88 mm Hg, 1.7 psia) reduced the heat flux by a factor of almost three [Fig. 5.4]. Stated another way, a low heat flux produced coefficients of 0.036 and 0.037, while a higher heat flux produced 0.033 and 0.035.

The author doubts the reliability of the inside coefficient obtained under vacuum conditions for two reasons:



27.22.2.5

errererse seessessi errererer suurvas seedererer vierri

Nu/Refore





First, reducing the flow rate through the tube lowers the heat flux and increases the inside convective thermal resistance. Since the condensing coefficient is inversely proportional to the heat flux raised to the onethird power, the outside condensing resistance decreases. Because of this, the inside resistance becomes the dominant factor in the overall thermal resistance of the tube. Any error in measuring the inside heat transfer coefficient will greatly amplify the computed error in the condensing resistance. Thus, the condensing coefficient is highly sensitive to the accuracy of the inside coefficient, particularly at low water velocities (and corresponding heat fluxes).

Second, the test apparatus was subject to radiofrequency (RF) interference from the power supply. As the power was cut back to lower the heat flux, the silicon controlled rectifier chopped the input signal proportionally and emitted the chopped portion as RF energy. Despite efforts to shield the cabling to the data-acquisition system, it still captured the RF signal with unacceptable results, and the thermopile was experiencing instantaneous fluctuations of several hundred microvolts. Increasing the power supply, however, decreased the errant signal until, at the maximum power input used for the high heat flux measurements, the thermopile showed a near-steady temperature difference that was within 0.1 °C of the quartz thermometer.

For these reasons, the Sieder-Tate coefficients obtained from the low heat-flux runs were discarded. Averaging the coefficients found from both measurements for the high heat flux case yielded a coefficient of 0.034 ± 0.001 .

B. FINNED TUBES

のとないという

Figure 5.5 shows a relative plot of the outside condensing coefficient for all six finned tubes compared with smooth tube results and the Nusselt line. Figure 5.6 shows the same results corrected for area increases due to finning. These data were computed using the leading coefficient on the inside of 0.034 obtained in the previous section.

All tubes tested show good enhancement over the smooth Notice the increase of the smooth tube over the tube. Nusselt line - a result of the inherent vapor shear in the test section. The increased scatter in the lower heat-flux range is again a function of the increased significance of errors in measuring the inside thermal resistances. The plotting program used neglected ordinate values below zero or greater than 10⁵, so the scattering in this portion of the plot is actually worse than it appears. Figure 5.7 is a plot of the values obtained for the outside heat transfer coefficients of the six finned tubes for a constant heat flux of 250,000 W/m² (79,000 Btu/hr-ft²) and a pressure of 88 mm Hg (1.7 psia). This plot more clearly shows the optimum pitch of 2.5 mm.

The heat-transfer characteristics for any tube will be enhanced by the addition of fins. In the case of purely convective heat transfer, the enhancement is a function of the increased surface area exposed to the fluid medium.

Filmwise condensation, however, is dissimilar in that the build-up of a condensate film acts as an additional thermal resistance for the heat-transfer process. The objective of tube enhancement, therefore, is to decrease this film thickness. The surface tension forces in the condensate tend to draw the liquid toward the fins, leaving the tube surface with a thinner film. The thinner film results in a higher heat-transfer coefficient. [Ref. 12].







STATISTICS.

入 う う う う つ

Outside condensing coefficients corrected for area. **Figure 5.6** a a sur a the state of the state of the survey of the s



SAMA SAMANA BANNAN SAMANAN MANANAN MANANAN ALAMANA

Contraction of the second



The optimum pitch found in this research increases the surface of the condenser tube 88%, but condensing heattransfer coefficients for that pitch were enhanced by as much as 330%.

The optimum pitch serves as the right trade-off between the attractive forces of the fins on the condensate and the channel area between the fins to drain the condensate from the tube. A pitch smaller than the optimum has too narrow a gap between fins to efficiently allow condensate run-off. This is because the condensate drawn to adjacent fins only combines to create a thicker film as shown in Figure 5.8.

Ň

277273 AUDIDIDI BUDDIDI KUMENE MELANG KUMAN

COMPANYA SONAFRA SUBSYSKA ARABARA



Smooth Tube



Pitch Too Small





VI. CONCLUSIONS AND RECOMMENDATIONS

A. CONCLUSIONS

1. The test apparatus successfully operates under vacuum conditions with no degrading effects caused by the presence of noncondensable gases.

2. The Sieder-Tate coefficient was determined to be 0.034 ± 0.001 using both an instrumented smooth tube and the modified Wilson plot technique.

3. An optimum fin pitch of 2.5 mm was found for the finned tubes tested.

4. Heat-transfer enhancements up to 300% were realized while the increase in the tube outer surface area (due to finning) was only 88% over the smooth tube.

B. RECOMMENDATIONS

1. Install a second, more reliable pressure transducer to replace the manometer.

2. Install a larger cooling-water sump to eliminate use of the once-through system from the fresh-water tap.

3. Collect data using an insert for the test tube to decrease the internal convective resistance and, thereby, decrease the uncertainty in the calculated condensing coefficient.

4. Take data for the finned tubes while varying fin height and thickness (as well as pitch).

<u>APPENDIX A</u> THERMOCOUPLE CALIBRATION

1. EQUIPMENT USED

addition sectors sectors

TOTAL CALL

a. Thermocouple Wire

Copper-constantan, 0.245-mm (0.01-in) Teflon-coated wire was used for all thermocouples.

b. Calibration bath

A Rosemont Engineering Model 913A calibration bath was used. A schematic representation of the bath is shown in Figure A.1.

1) Heating: Electrical

2) Cooling: Liquid Nitrogen

Note: Once a desired temperature is reached, the temperature is held constant by rapid cycling between heating and cooling. The bath is rated for temperature fluctuations of less than 0.01°C.

c. Thermocouple readout

An HP 3054A Data Acquisition/Control System was used to obtain data. Resolution of the acquisition system was $1 \not V$.

d. Bath temperature measurement

A platinum resistance thermometer with an accuracy of 0.01 C was used.

2. PREPARATION

a. Procedure

The instrumented tube (with the wall thermccouples installed) and the steam thermocouples were immersed in the bath as well as the probes for the quartz thermometer.

b. Analysis

The computer program TCAL was used to monitor and store all thermocouple readings on a disk. A listing of the program is located in Appendix E. 3. CALIBRATION PROCEDURE

a. The bath temperature was set at about 10 °C.

b. When the bath temperature reached steady state, its value was entered into the computer.

c. The computer automatically recorded and printed all thermocouple readings.

d. The bath temperature was raised in increments of 10
 C to 90 °C and steps b and c were repeated for each increment.

4. CALIBRATION CURVES

a. A least-squares method was used to generate a polynomial of the form:

 $D_{T} = a_{o} + a_{T} + a_{z}T^{2}$

where: D_{\uparrow} is the difference between the bath and thermocouple temperatures, and

T is the value of the thermocouples obtained using the seventh-order polynomial fitted for the Type T thermocouple wire used. This polynomial is listed in the beginning of the program TCAL.

b. Coefficient values obtained were:

 $a_o = 4.7338 \times 10^{-3}$ $a_i = 7.6928 \times 10^{-3}$

 $a_{1} = -8.0779 \times 10^{-5}$

c. A plot of the curves is found in Figure A.1. The curve for the wall thermocouples gives a different reading due to the thermal conduction of the air temperature through the tube. The data used for calibration did not include this average value. Only the value obtained for the two steam thermocouples was used for all of the thermocouples.





APPENDIX B SAMPLE CALCULATIONS

1. HEAT LOSS

うたいかい

A set of calculations was performed to estimate the heat loss of the apparatus due to natural convection. These calculations were performed for two main sections: the boiler (with and without insulation) and the piping to the test section. The minimum power input used during the thesis was 17 kW.

a. Boiler (without insulation)

1) Given the following dimensions and properties of the toiler section:

 $r_{i \circ \beta} = 0.1524 \text{ m}$ $r_{c \circ \theta} = 0.1561 \text{ m}$ $L_{\beta} = 0.4064 \text{ m}$ $k_{c} = 1.4 \text{ W/m-K}$

2) Given the following temperatures:

 $T_{i,B} = 100^{\circ}C$ $T_{o,B} = 95^{\circ}C$ $T_{\infty} = 25^{\circ}C$

 $T_{f} = (95+25)/2 = 60 \,^{\circ}C$

3) Given the following properties of air at the film temperature:

 $k_A = 28.74 \times 10^{-3} W/m-K$

Pr = 0.702 $r = 19.21 \times 10^{-6} \text{ m}^2/\text{s}$ (3 = 0.003 K⁻¹

4) Compute the Grashof Number for the boiler assuming laminar conditions ($Ra < 10^9$):

 $Gr_{B} = g(T_{o,B} - T_{-}) L^{3}/\gamma^{2}$ $Gr_{3} = 3.7 \times 10^{8}$

5) Compute the Rayleigh Number for the boiler: $Ra_{B} = Gr Pr$ $Ra_{B} = 2.6x10^{8}$ (< 10⁹)

6) Determine the applicability of a flat-plate analysis according to the method of Sparrow and Gregg [Ref. 13]

 $L / r_{o,8} < (0.15/\sqrt{8}) Gr^{1/4}$

2.60 < 7.36

というない

8) Compute the average Nusselt Number (via flat-plate analysis):

 $\overline{Nu} = [0.825 + 0.387 Ra_{g}^{1/6} / [1 + (0.492/Pr)^{9/16}]^{8/27}]^{2}$

 $\overline{Nu} = 81.3$

9) Compute the external heat-transfer coefficient:

 $\bar{h}_o = N\bar{u} k / L$

 $\bar{h}_{\rho} = 5.75 \text{ W/m}^2 - \text{K}$

10) Compute the external convective thermal resistance:

$$R = 1/(h_0 2 \pi r_{0,3} L_B)$$

R = 0.436 K/W

11) Calculate the wall resistance:

$$R_{w} = \ln (T_{v,\theta}/T_{i,\theta})/2\pi k_{c} L_{\theta}$$

$$R_{w} = 0.007 \text{ K/W}$$
12) Neglecting the wall resistance, calculate the heat loss from the boiler:

$$Q_{\theta} = (T_{i,\theta}-T_{\star})/(R_{\star}+R_{w})$$

$$Q_{\theta} = 169.3 \text{ W}$$
b. Boiler (with insulation)

$$Q_{\theta}' = 29.2 \text{ W}$$
c. Piping

$$Q_{p} = 65.0 \text{ W}$$
d. Total (worst case)

$$Q = Q_{\theta} + Q_{P}$$

$$Q = 234.3 \text{ W} (<< 17000 \text{ W of input power})$$

a tradector texesters territies territies territies territies

····

<u>APPENDIX C</u> STEAM-CLEANING PROCEDURE

Note: The procedures listed here provide an excellent method of cleaning the apparatus, but will cause the walls of the apparatus to heat to temperatures over 100 C (212 F). Once this happens, several hours are required before the walls can cool down such that they won't superheat the steam generated during an experimental run. Therefore, use this method only prior to operating at atmospheric pressure or when the system is badly contaminated.

1. Ensure the water level in the boiler is at least six inches above the upper ends of the heaters.

2. Remove the thermopile from the inlet side of the test section.

3. Energize the power supply and adjust the rheostat until the voltmeter reads about 100 V.

4. Close the recirculating valve.

5. Open the fill/drain valve.

6. Once vapor begins exiting the fill/drain line, allow the system to steam for several minutes.

7. To maximize the steam flow through the test section, fully open the throttle to the auxiliary condenser.

8. When done, simultaneously close the drain/fill valve and open the recirculating valve.

9. Circulate cooling water through the test tube and check for the presence of any dropwise condensation.

10. Reinstall the thermopile.

APPENDIX D

SYSTEM START-UP AND SHUT-DOWN PROCEDURES

To start the system:

1. Fill the boiler to at least six inches above the upper level of the heaters.

2. Energize the air ejector for ten minutes.

3. Circulate water through the cooling water sump by opening the inlet value and activating the siphon.

4. Energize the power to the boiler and adjust the rheostat until the voltmeter reads 90 V for vacuum operation, or 170 V for atmospheric pressures.

5. Energize the circulating pumps and adjust the throttle for the desired flow through the tube.

6. Open the value to the auxiliary condenser and adjust the flow: about 15% for vacuum operation or 30% for atmospheric runs.

7. Observe the steam temperature indicated on th efront of the data acquisition unit until the voltage is reached corresponding to the saturation temperature of the the desired operating pressure.

8. Energize the air ejector for two minutes.

9. Adjust the the flow rate through the auxiliary condenser to bring the apparatus to the desired operating pressure.

To secure the system:

- 1. Secure cooling water to the test tube.
- 2. Secure power to the boiler.
- 3. Open the vent valve.
- 4. Secure the water supply to the auxiliary condenser.

APPENDIX B Computer program Listings

The following pages contain the computer program listings used for this thesis:

SIEDER (page 61) WILSON (page 65) DRP (page 71) ICAL (page 80)

a dirent by the total

PROGRAM SIEDER

OT AND ADDRESS OF ADDRESS ADDR

うちま あいまい

LANGER AND

1000! FILE NAME: SIL 1010! DISK NUMBER: 12 SIEDER November 29, 1983 1020! REVISED: 1030! COM /Cc/ C(7) DIM Emf(10), Tw(5) 1050 DATA 0.10086091.25727.94369.-757345.8295.78025595.81 1060 1070 DATA -9247486589.6.97688E11,-2.6619213.3.94078E14 READ C(+) 1080 Kcu=385 1090 1100 Di=.0127 Do=.01905 1110 1120 Dr=.015875 1130 L=.13335 1140 L1=.060325 L2=.034925 1150 PRINTER IS 701 1160 1170 BEEP CLEAR 709 INPUT "ENTER MONTH. DATE AND TIME (MM:DD:HH:MM:SS)",BS OUTPUT 709;"TD";BS 1180 1190 1200 1210 1220 1220 1230 1240 Series:! OUTPUT 709:"TD" ENTER 709:A\$ PRINT USING "10X,""Month, date and time: "",14A";AS 1250 1260 1270 BEEP INPUT "ENTER DISK NUMBER", Dn PRINT PRINT USING "10X,""NOTE: Program name: SIEDER""" PRINT USING "16X.""Disk number = "",DD";Dn 1280 1290 1300 BEEP INPUT "ENTER INPUT MODE (1=3054A.2=FILE)", Im 1310 IF Im=1 THEN 1320 BEEP 1330 1340 1350 INPUT "GIVE A NAME FOR THE DATA FILE".D_file\$ CREATE BDAT D_file\$,10 1360 ELSE 1370 BEEP INPUT "GIVE THE NAME OF THE DATA FILE", D_files 1380 BEEP 1390 INPUT "ENTER THE NUMBER OF RUNS STORED",Nrun PRINT_USING "16X,""This analysis was performed for file "",10A";D_file\$ 1400 1410 1420 END IF BEEP 1430 INPUT "GIVE A NAME FOR PLOT DATA FILE", Plots 1440 1450 BEEP INPUT "ENTER OPTION FOR END-FIN EFFECT (1=Y,0=N)".Ife IF Ife=0 THEN PRINT USING "16X,""This analysis neglects end-fin effect""" IF Ife=1 THEN PRINT USING "16X.""This analysis includes end-fin effect""" 1460 1470 1480 CREATE BDAT Plot\$,5 ASSIGN @File TO D_file\$ ASSIGN @Filep TO Plot\$ 1490 1500 1510 1520 J=0 1530 1540 1550 Sx=0 Sy=0 Sxs=0 1560 Sxy=0 IF Im=1 THEN 1580! READ DATA THROUGH THE DATA ACQUISITION SYSTEM

÷...

M

1590! IF THE INPUT MODE (Im) = 1 1600 BEEP BEEP INPUT "ENTER FLOWMETER READING".Fm OUTPUT 709:"AR AF20 AL30 VR1" FOR I=0 TO 10 OUTPUT 709:"AS SA" IF I>4 THEN 1610 1620 1630 1640 1650 1660 1660 Se=0 FOR K=0 TO 19 ENTER 709;E 1680 Se=Se+E NEXT K 1690 1700 1710 Emf(I)=ABS(Se/20) 1720 ELSE 1730 1740 1750 ENTER 709:E Emf(I)=ABS(E) END IF NEXT I OUTPUT 713;"T1R2E" 1760 1770 WAIT 2 ENTER 713;T11 OUTPUT 713;"T2R2E" .1780 1790 1800 WAIT 2 ENTER 713:T2 OUTPUT 713;"T1R2E" 1810 1820 1830 WAIT 2 ENTER 713:T12 T1=(T11+T12)*.5 1840 1850 1860 ELSE 1870 READ DATA FROM A USER-SPECIFIED FILE IF INPUT MODE (Im) = 2 ENTER_#File;Emf(*),T1.T2.Fm 1880! 1890! 1900 1910 1920 END IF Tavg=(T1+T2)*.5 Twal1=0 FOR I=5 TO 10 Tw(I=5)=FNTvsv(Emf(I)) 1930 1940 1950 1970 Twall=Twall+Tw(I-5) NEXT I Twall=Twall/6 1980 1990 Twall-Twall/0 Cpw=FNCpw(Tavg) Rhow=FNRhow(Tavg) Hd=5.00049E-3+6.9861937E-3+Fm Md=Nd=(1.0365-1.96644E-3+T1+5.252E-6+T1^2)/.995434 2000 2010 2020 2030 Hf=Md/Rhow 2040 Mt = Md/know Vw=Mf/(PI=Di^2/4) T2c=T2-(.0138+.001=Vw^2) T2c=T2-.004=Vw^2 Q=Md=Cpw=(T2c-T1) 2050 2051 2060! 2070 2080 Dtw-Q+LOG(Do/Di)/(2+PI+Kcu+L)+.5 Tuall=Tuall-Dtu 2090 Lmtd=(T2c-T1)/LOG((Twall-T1)/(Twall-T2c)) 2100 2110 2120 2130 Kw=FNKw(Tavg) P1=PI=(Di+Do) P2=PI=(Di+Dr) 2130 2140 2150 2160 2170 2180 Al=(Do-Di)=PI=Do A2=(Dr-Di)=PI=Dr Hi=Q/(PI=Di=L=Lmtd) IF Ife-0 THEN Hic-Hi GOTO 2300 2190

SUSSES SUCCESSES SCREEKE SUGGESTER SUSSESSES SUSSESSES SUSSESSES

このできるです。 スペルノシノノ

いいい 副子 いいい

2200 END IF M1=(Hi+P1/(Kcu+A1))`.5 M2=(Hi+P2/(Kcu+A2))^.5 2210 2220 2230 Fe1=FNTanh(M1+L1)/(M1+L1) 2240 2250 Fe2=FNTanh(M2=L2)/(M2=L2) Hic=Q/(PI+Di+(L+L1+Fe1+L2+Fe2)+Lmtd) 2260 IF ABS((Hi-Hic)/Hic)>.01 THEN 2270 2280 Hi=(Hic+Hi)*.5GOTD 2210 2290 END IF 9 PRINT USING "10X,""Position number 6""" 2300 2 3 2310 1 5 . 2320 PRINT USING "10X.""Wall temperature (Deg C) : "",6(DD.DD,1X)":Tw(*) 2330! CALCULATE THE NUSSELT NUMBER Nu=Hic+Di/Kw 2340 2350 Muw=FNMuw(Tavg) 2360 Re=Rhow=Vw=Di/Muw Cf=(Muw/FNMuw(Twall))^.14 2370 2370 Cf=(nuw/FNnuw(iwall)), 14 2380 Prw=FNPrw(Tavg) 2390 X=Re^.8*Prw^.3333*Cf 2400! COMPUTE COEFFICIENTS FOR THE LEAST-SQUARES-FIT 2410! STRAIGHT LINE 2420 OUTPUT @Filep:X.Nu 2430 PRINT USING "10X,""Twall Tin Tout Lmtd Vw X Nu""" 2440 PRINT USING "10X,4(2D.2D.2X),Z.DD,2X.5D.D,2X.4D.D";Twall,T1,T2c,Lmtd,Vw.X, Ňц 2450 Sx=Sx+X 2460 Sy=Sy+Nu 2470 Sxs=Sxs+X+X Sxy=Sxy+X+Nu STORE RAW DATA IN A USER-SPECIFIED FILE IF INPUT MODE (Im) = 1 2480 2490! 2500! IF Im=1 THEN OUTPUT @File:Emf(=),T1.T2.Fm BEEP 2510 2520 J=J+1 IF Im=1 THEN INPUT "ARE YOU TAKING MORE DATA (1=YES.0=N0)?",Go_on 2530 2540 2550 2560 Nrun=J IF Go_on=1 THEN 1570 2570 2580 IF J<Nrun THEN 1570 END IF 2590 2600 2610! Ci=Sxy/Sxs Ci=(Nrun*Sxy-Sy*Sx)/(Nrun*Sxs-Sx`2) 2620 2630 Ac=(Sy-Ci#Sx)/Nrun PRINT 2640 PRINT USING "10X.""Sieder-Tate Coefficient = "",D.4D";Ci 2650 PRINT 2660 PRINT USING "10X.""Least-Squares Line:""" PRINT USING "12X.""Slope - "".MD.5 PRINT USING "12X.""Intercept - "".MD.5 2670 - "", MD.SDE.";Ci - "", MD.SDE,";Ci 2680 2690 2691 PRINT 2700 IF Im=1 THEN 2720 BEEP PRINT USING "10X,""NOTE: "",ZZ."" data runs were stored in file "",8A";Nru 2730 n.D_file\$ 2731 ELS ELSE 2732 PRINT USING "10X,""NOTE: The above analysis was performed for file "".14A" ;D_fileS

63

2740 2750 2760 2770 2780 2790 END IF PRINT USING "16X.""Plot data are stored in file "",14A":Plot\$ ASSIGN @File TO * ASSIGN @Filep TO * END END DEF FNRhow(T) Ro=1006.35724-T*(.774489-T*(2.262459E-2-T*3.03304E-4)) RETURN Ro 2790 2800 2810 2820 2830 2830 2840 2850 2850 RETURN Prw RETURN Prw RETURN Prw RETURN Prw FNEND PNEND DEF FNMuw(T) A=247.8/(T+133.15) Muw=2.4E-5=10^A RETURN Muw 2870 2880 2890 2900 2910 2920 2930 2940 2950 2950 2960 2970 2980 FNEND DEF FNKw(T) Kw=.572183504477+1.52770121209E-3*T RETURN Kw FNEND DEF FNTvsv(Emf) COM /Cc/ C(7) Sum=C(0) FOR I=1 TO 7 2990 3000 3010 3020 3030 Sum=Sum+C(I)*Emf^I NEXT I RETURN Sum FNEND DEF FNCpw(T) Cpw=(4.21120858-T*(2.26826E-3-T*(4.42361E-5+T*2.71428E-7)))*1000 RETURN Cpw 3040 3050 3060 3070 3080 3090 FNEND DEF FNTanh(X) P=EXP(X) 3100 3110 3120 Q=EXP(-X) Tanh=(P+Q)/(P-Q) RETURN Tanh 3130 FNEND

100000

PROGRAM WILSON

SECOND MARCHINE

12022001

1000! FILE NAME: WILSON 1010! REVISED: Decemb December 5. 1983 1020! COM /Cc/ C(7) DATA 0.10086091.25727.94369.-767345.3295.78025595.81 DATA -9247486589.6.97688E11.-2.66192E13.3.94078E14 1030 1040 1050 READ C(+) 1060 1070 DIM Emf(4) 1080 L=.130175 1090 L1-.060325 1100 L2=.034925 1110 Do=.01905 1120 Di=.0127 1130 Dr=.015785 1140 Kcu=385 1150 Rm=Do+LOG(Do/Di)/(2+Kcu) PRINTER IS 701 1160 1170 BEEP CLEAR 709 INPUT "ENTER MONTH. DATE, AND TIME (MM:DD:HH:MM:SS".BS 1180 1190 1200 1210 1220 1230 1240 OUTPUT 709; "TD"; B\$ Jp=0 OUTPUT 709:"TD" ENTER 709:AS PRINT USING "10X,""Month, date and time : "",14A":AS 1250 1260 1270 BEEP INPUT "ENTER DISK NUMBER", Dn PRINT 1280 1290 PRINT USING "10X.""NOTE: Program name : M_H PRINT USING "16X,""Disk number = "",DD";Dn : M_WILSON""" 1300 BEEP 1310 1320 1330 INPUT "ENTER INPUT MODE (1=3054A,2=FILE)", Im IF Im=1 THEN BEEP INPUT "GIVE A NAME FOR THE DATA FILE", D_files CREATE BDAT D_files, 10 1340 1350 1360 ELSE 1370 BEEP INPUT "GIVE THE NAME OF THE DATA FILE",D_file\$ PRINT USING "16X,""This analysis is for data in file "",14A";D_file\$ 1380 1390 1400 BEEP INPUT "ENTER THE NUMBER OF RUNS STORED", Nrun 1410 END IF 1420 1430 BEEP 1440 1450 INPUT "GIVE A NAME FOR PLOT-DATA FILE".Plots BEEP 1460 INPUT "ENTER OPTION (1=QCT,2=T-PILE,3=AVE)", Itm 1480 INPUT "ENTER OPTION FOR END-FIN EFFECT (1=Y,0=N)".Ife 1480 IF Itm=1 THEN PRINT USING "16X,""This analysis uses QCT readings""" 1500 IF Itm=2 THEN PRINT USING "16X,""This analysis uses I-PILE readings""" 1510 IF Itm=3 THEN PRINT USING "16X,""This analysis uses average of QCT and I-P ILE readings""" 1520 IF Itm=1 THEN POINT USING "16X," IF Ife=1 THEN PRINT USING "16X,""This analysis includes end-fin effect""" IF Ife=0 THEN PRINT USING "16X,""This analysis neglects end-fin effect""" CREATE BDAT Plot\$,10 ASSIGN @Filep TO Plot\$ 1520 1530 1540 1550 1560! Ciu=.040 1570! Cil=.028

J]=0 Ci=.03 J=0 1580 1590 1600 1610! Ci=(Ciu+Cil)*.5 1620 Sx=0 1630 Sy=0 1640 1650 1660 Sxs=0 Sxy=0 PRINT PRINT USING "10X,""Iteration number 1670 f+1L;"00,"" = 1680 IF JJ=0 OR Jp=1 THEN 1690 PRINT The LND IF The TO D_{file} To T_{T2} The Lmt T2 To T_{T2} ASSIGN F_{File} TO D_{file} 1730 IF Im=1 AND Jj=0 THEN 1740! READ DATA THROUGH THE DATA ACQUISITION SYSTEM 1750! IF THE INPUT MODE (Im) = 1 1760 BEEP 1770 THEORY Х Lmtd Via 1760 INPUT "ENTER FLOWMETER READING",Fm OUTPUT 709:"AR AF60 AL63" OUTPUT 709:"AS SA" 1780 1790 1800 Etp=0 1810 1820 1830 FOR I=1 TO 20 ENTER 709;Et Etp=Etp+Et NEXT I 1840 Etp=Etp/20 OUTPUT 709;"AS SA" 1850 1860 Ptran=0 FOR I=1 TO 50 ENTER 709;Pt 1870 1880 1890 1900 Ptran=Ptran+Pt 1910 NEXT I Ptran=Ptran/50 OUTPUT 709:"AS SA" ENTER 709:Bvol OUTPUT 709:"AS SA" ENTER 709:Bamp OUTPUT 709:"AR AF20 AL24" EDD 10 4 1920 1930 1940 1950 1960 1970 FOR I=0 TO 4 OUTPUT 709:"AS SA" ENTER 709:Emf(I) 1980 1990 2000 Emf(I)=ABS(Emf(I)) NEXT_I 2010 2020 OUTPUT 713:"T1R2E" 2030 2040 2050 WAIT 2 ENTER 713:T11 DUTPUT 713:"T2R2E" 2060 2070 HAIT 2 HAI1 2 ENTER 713:T2 OUTPUT 713:"T1R2E" HAIT 2 ENTER 713:T12 T1=(T11+T12)=.5 2080 2090 2100 2110 2120 2130 2140 CLEAR 713 ELSE 21501 READ DATA FROM A USER-SPECIFIED FILE IF INPUT MODE (Im) = 2 ENTER OFile; Bvol, Bamp.Ptran, Etp.Emf(*), Fm, T1, T2 END IF 2160 2170 2180 Tsat=FNTvsv((Emf(0)+Emf(1))+.5)

25221 20222223, 20222223 SMANSSON ANALAUG 2020000

1000 (NO000000)

Ş

Ti=FNTvsv(Emf(2)) 2190 22190 22200 22210 2220 2230 2240 2250 2250 2250 2260 Grad=FNGrad((T1+T2)*.5) To=Ti+ABS(Etp)/(10+Grad)+1.E+6 IF JI=0 THEN Er1=ABS(TI-T1) PRINTER IS 1 PRINT USING """T1 PRINT USING """T1 - "".DD.3D":T1 - "".DD.DD":Ti IF Er1>.5 THEN 2280 2290 2300 BEEP PRINT "QCT AND TC DIFFER MORE THAN 0.5 C" BEEP 2310 2320 2330 INPUT "OK TO GO AHEAD (1=Y,0=N)?", Ok1 INPUT OK TO GO HHEHD (1=1,0=N)/",021 END IF PRINT USING """DT (QCT) = "",Z.3D";T2-T1 PRINT USING """DT (T-PILE) = "",Z.3D";T0-T1 IF Ok 1=0 AND Ert>.5 THEN 3600 Er2=ABS((T2-T1)-(T0-T1))/(T2-T1) IF Er2>.05 THEN 2340 2350 2360 2370 BEEP 2380 2390 PRINT "QCT AND T-PILE DIFFER MORE THAN 5%" 2400 BEEP INPUT "OK TO GO AHEAD (1=Y,0=N)?",0k2 IF 0k2=0 AND Er2>.05 THEN 3600 2410 2420 END IF 2430 2440 2450 PRINTER IS 701 2450 END IF 2450! CALCULATE THE LOG-MEAN-TEMPERATURE DIFFERENCE 2470 IF Itm=1 THEN 2480 Tf=T1 2490 T1=T2 2500 2510 END IF IF Itm=2 THEN 2520 2530 Tf=Ti T1=To END IF IF Itm=3 THEN TF=(T1+T1)=.5 2540 2550 2560 2570 T1=(T2+To)+.5 2580 2590 END IF Tavg=(Tf+TL)+.5 Trise=T1-Tf Lmtd=Trise/LOG((Tsat-Tf)/(Tsat-T1)) Cpw=FNCpw(Tavg) 2600 2610 2620 2630 2640 2650 2660 2670 Rhow=FNRhow(Tavg) Ku=FNKu(Tavg) Muwa=FNMuw(Tavg) Prw=FNPrw(Tavg) Mdt=5.00049E-3+6.9861937E-3*Fm Md=Mdt+(1.0365-Tf+(1.96644E-3-Tf+5.252E-6))/.995434 2680 Vf=Md/Rhow 2690 Vw=Vf/(PI+Di^2/4) 2700 2710 2720 2730 2740 2750 Trise Trise .004+Vw^2 Q=Md+Cpw=Trise Qp=Q/(PI+Do+L) Uo=Qp/Lmtd Re=Rhow+Vu+Di/Muwa 2760 2770 Fe1=0 Fe2=0 2780 Cf=1 2790 Two=Tsat-5

2800 Tfilm=Tsat/2+Two/2 2810 Kf=FNKw(Tfilm) 2820 Rhof=FNRhow(Tfilm) 2830 Muf=FNMuw(Tfilm) Hfgp=FNHfg(Tsat)+.68*FNCpw(Tfilm)*(Tsat-Two) New=Kf*(Rhof^2*9.799*Hfgp/(Muf*Do*Gp))^.3333 2840 2850 2860 2870 Ho=.655+New Twoc=Tsat-Qp/Ho 2880 IF ABS((Twoc-Two)/Twoc)>.001 THEN 2890 Two=Twoc GDTO 2800 2900 2910 END IF 2920 Cf=1.0 Chega=Re².8*Prw³.3333*Cf Hi=Kw/Di*Ci*Omega IF Ife=0 THEN 3040 P1=PI*(Di+Do) 2930 2940 2950 2960 2970 P2=PI*(Di+Dr) A1=(Do-Di)=PI=(Di+Do)=.5 A2=(Dr-Di)=PI=(Di+Dr)=.5 2980 2990 M1=(Hi=P1/(Kcu=A1))^.5 M2=(Hi=P2/(Kcu=A2))^.5 3000 3010 3020 Fe1=FNTanh(M1+L1)/(M1+L1) 3030 3040 3050 Fe2=FNTanh(M2*L2)/(M2*L2) Dt=Q/(PI*Di*(L+L1*Fe1+L2*Fe2)*Hi) Cfc=(Muwa/FNMuw(Tavg+Dt))^.14 3060 3070 IF ABS((Cfc-Cf)/Cfc)>.01 THEN Cf=(Cf+Cfc)+.5 GOTO 2930 JXS=SXS+X=X J170 Sxy=Sxy+X=Y 3180! STORE RAW DATA IN A USER-SPECIFIED FILE IF INPUT MODE (Im) = 1 3190 IF Im=1 AND Jj=0 THEN OUTPUT @File:Bvol.Bamp.Ptran.Etp.Emf(*),Fm.T1.T2 3200 IF Jj=0 OR Jp=1 THEN PRINT USING "8X.5(2X.3D.DD),2(2X.D.5D)";Tf,T1.Tsat.Lm td.Vw.X.Y 3210 BEEP 3220 J=J+1 3230 IF Im=1 AND Jj=0 THEN 3240 INPUT "DO YOU HAVE MORE DATA (1=Y.0=N)?",Go on 3250 Nrun=J 3260 IF Go_on=1 THEN 1730 3270 ELSE 3280 IF JCNrun THEN 3290 END TT 3080 X=Do+New/(Omega+Kw) Y=New+(1/Uo-Rm) COMPUTE CDEFFICIENTS FOR THE LEAST-SQUARES-FIT STRAIGHT LINE IF Jp=1 THEN OUTPUT @Filep:X.Y Sx=Sx+X 3090 END IF Sl=(Nrun=Sxy-Sy=Sx)/(Nrun=Sxs-Sx`2) Ac=(Sy-Sl=Sx)/Nrun Cic=1/Sl 3290 3300 3310 3320 3330 Jj=Jj+1 IF Jp=1 THEN Jp=2 IF ABS((Cic-Ci)/Cic)>.001 THEN Ci=(Cic+Ci)=.5 PRINT USING "10X.""Intermediate Sieder-Tate coefft = "".Z.4D":Ci 3340 3350 3360 3370 GOTO 1600 3380

i.

1000000

วงรบ 3400 IF Jp=0 THEN Jp=1 END IF 3410 3420 IF Jp=1 THEN 1600 Ci=(Ci+Cic)*.5 PRINT 3430 3440 = "".Z.4D":C1 PRINT USING "10X,""Sieder-Tate coefficient 3450 3460 PRINT PRINT USING "10X.""Least-Squares Line:""" PRINT USING "10X."" Slope = "".Z.5DE.";Sl PRINT USING "10X."" Intercept = "",MZ.5DE,";Ac 3470 3480 3490 3500 PRINT IF Im=1 THEN BEEP 3510 3520 3530 PRINT USING "10X,""NOTE: "",ZZ."" data runs are stored in file "",8A";J.D_ file\$ 3540 ELSE 3540 3550 PRINI . 10A":D_file\$ 2560 END IF PRINT USING "10X.""NOTE: Above analysis was performed for data in file "", 3570 PRINT USING "16X,""Plot data are stored in file "",10A";PlotS 3580 ASSIGN @File TO * ASSIGN @Filep TD * 3590 3600 END 3610 3620 3630 DEF FNRhow(T) Ro=1006.35724-T+(.774489-T+(2.262459E-2-T+3.03304E-4)) RETURN Ro 3640 FNEND DEF FNPrw(T) Prw=FNCpw(T)*FNMuw(T)/FNKw(T) RETURN Prw 3650 3660 3670 3680 FNEND DEF FNMum(T) A=247.8/(T+133.15) Mu=2.4E-5+10^A 3690 3700 3710 3720 3730 RETURN ML FNEND 3740 DEF FNKw(T) 3750 3760 X=(T+273.15)/273.15 Ku=-.92247+X=(2.8395-X=(1.8007-X=(.52577-.07344=X))) RETURN KW 3770 FNEND DEF FNTvsv(Emf) COM /Cc/ C(7) 3780 3790 3800 Sum=C(0) FOR I=1 TO 7 3810 3820 3830 Sum=Sum+C(I)+Emf^I NEXT I RETURN Sum 3840 3850 3860 FNEND 3870 DEF FNCpw(T) 3880 Cpw=(4.21120858-T+(2.26826E-3-T+(4.42361E-5+2.71428E-7)))+1000 RETURN Cpu 3890 3900 FNEND 3910 DEF FNTanh(X) 3920 P-EXP(X) Q=EXP(-X) 3930 3940 Tanh=(P+Q)/(P-Q)3950 RETURN Tanh 3960 FNEND 3970 DEF FNGrad(T)

WALLERED, WEATHER READERS STRUCTURE TOTATION

CARLEN AND AND AND ALLEN
3980 COM /Cc/ C(7) 3990 Grad=37.9853+.104388+T 4000 RETURN Grad 4010 FNEND 4020 DEF FNFvst(T) 4030 F=466.444+T*(7.09451-T*1.65808E-2) 4040 RETURN F 4050 FNEND 4060 DEF FNHfg(T) 4070 Hfg=2477200-2450*(T-10) 4080 RETURN Hfg 4090 FNEND

X WIND

大いのないない ないたいたいし うちゃくろう

PROGRAM DRP

ANAL MANANAN MALIMAR VALAMMA MANANANA ANALAMA

```
1000! FILE NAME: DRP
1010! REVISED: November 18. 1983
1020!
        COM /Cc/ C(7)
DIM Emf(10)
1030
1040
        DATA 0.10086091.25727.94369.-767345.8295.78025595.81
        DATA -9247486589,6.97688E+11.-2.66192E+13.3.94078E+14
1060
        READ C(+)
1070
1080
        Di=.0127
                           ! Inside diameter of test tube
                          ! Outside diameter of test tube
! Outside diameter of the outlet end
1090
        Do=.01905
        Dr=.015875
1100
1110
        Dssp=.1524
                           ! Inside diameter of stainless steel test section
1120
1130
        Ax=PI=Dssp 2/4-PI=Do+L
                          ! Condensing length
! Inlet end "fin length"
        L=.130175
·1140
        L1=.060325
                          ! Dutlet end "fin length"
! Thermal conductivity of Copper
1150
1160
        L2=.034925
        Ci=.034 ! Sieder-Tate coefficient
Rm=Do+LOG(Do/Di)/(2*Kcu) ! Wall resistance based on outside area
PRINTER IS 701
1170
1180
 1190
        CLEAR 709
1200
1210
         BEEP
        INPUT "ENTER MONTH, DATE AND TIME (MM:DD:HH:MM:SS)",Dates
OUTPUT 709:"TD";Dates
OUTPUT 709:"TD"
1220
1230
1240 1250
        ENTER 709:Date$
1260
1270
                                Month. date and time :";Date$
        PRINT
         PRINT
1280
        PRINT USING "10X,""NOTE: Program name : DRP"""
 1290
         BEEP
         INPUT "ENTER DISK NUMBER".Dn /
PRINT USING "16X,""Disk number = "",DD";Dn
 1300
 1310
 1320
         BEEP
 1330
         INPUT "ENTER INPUT MODE (1=3054A,2=FILE)", Im
 1340
         IF Im=1 THEN
 1350
         BEEP
        INPUT "GIVE A NAME FOR THE RAW DATA FILE".D_file$
CREATE BDAT D_file$.15
ASSIGN @File TO D_file$
 1360
 1370
1380
1390
         BEEP
         INPUT "ENTER GEOMETRY CODE (1-FINNED.0-PLAIN)",Ifg
OUTPUT OFile:Ifg
IF_Ifg=0 THEN
 1400
 1410
 1420
         BEEP
 1430
 1440
         INPUT "WALL TEMPERATURE MEASUREMENT (1=Y, 0=N)", Iwt
 1450
         ELSE
         BEEP
 1460
         INPUT "ENTER FIN PITCH, WIDTH AND HEIGHT", Fp.Fw.Fh
END IF
 1470
 1480
         IF Ifg=0 THEN OUTPUT @File;Iwt
IF Ifg=1 THEN OUTPUT @File:Fp,Fw,Fh
 1490
 1500
 1510
         ELSE
 1520
1530
         BEEP
         INPUT "GIVE THE NAME OF THE EXISTING DATA FILE",D_file$
PRINT USING "16X,""This analysis was performed for data in file "".10A":D_
 1540
```

3

1162 1550 BEEP 1560 INPUT "ENTER THE NUMBER OF RUNS STORED". Nrun ASSIGN @File TO D_fileS 1570 ENTER OFILE IN U_tileS ENTER OFILE:Ifg IF Ifg=0 THEN ENTER OFILE:Iwt IF Ifg=1 THEN ENTER OFILE:Fp,Fw,Fh END IF 1580 1590 1600 1610 1620 IF Ifg=0 THEN 1630 BEEP 1640 1650 INPUT "WANT TO CREATE A FILE FOR Nr vs F (1=Y,0=N)?", Inf ELSE 1660 Inf=0 END IF 1670 1680 IF Inf=1 THEN 1690 BEEP INPUT "GIVE A NAME FOR Nr vs F FILE" .Nrfs 1700 CREATE BDAT Nrf\$.2 1710 1720 ASSIGN ONrf TO Nrfs END IF BEEP 1740 1750 1760 1770 INPUT "ENTER OPTION (1=QCT,2=T-PILE,3=AVE)", Itm BEEP 1760 BEEP 1770 INPUT "ENTER OPTION FOR END-FIN EFFECT (I=Y.O=N)".Ife 1780 IF Itm=1 THEN PRINT USING "16X,""This analysis uses QCT readings""" 1790 IF Itm=2 THEN PRINT USING "16X,""This analysis uses T-PILE readings""" 1800 IF Itm=3 THEN PRINT USING "16X.""This analysis uses average of QCT and T-P ILE readings""" 1900 IF Itm=3 THEN PRINT USING "16X.""This analysis uses average of QCT and T-P ILE readings"" IF Ife=1 THEN PRINT USING "16X,""This analysis includes end-fin effect""" IF Ife=0 THEN PRINT USING "16X,""This analysis neglects end-fin effect""" PRINT USING "16X,""Sieder-Tate coefficient = "",Z.4D";Ci 1810 1820 1830 1840 BEEP INPUT "GIVE A NAME FOR PLOT DATA FILE",P_file\$ CREATE BDAT P_file\$,5 ASSIGN @Filep TO P_file\$ 1850 1860 1870 IF Iwt=1 THEN BEEP 1880 1890 INPUT "GIVE A NAME FOR WALL TEMPERATURE FILE",Wtfs CREATE BDAT Wtfs,5 ASSIGN @File1 TO Wtfs 1900 1910 1920 1930 END IF BEEP 1940 1950 INPUT "ENTER DUTPUT VERSION (1-SHORT,2-LONG)". Iov 1960! IF Im=1 THEN 1970! OUTPUT SFile:Ifg 1980! IF Ifg=0 THEN DUTPUT @File:Iwt 1990! IF Ifg=1 THEN DUTPUT @File:Fp.Fw.Fh 2000! ĒLSĒ 2010! ELSE 2010! ENTER #File:Ifg 2020! IF Ifg=0 THEN ENTER #File:Iwt 2030! IF Ifg=1 THEN ENTER #File:Fp,Fw,Fh 2040! END IF 2050 IF Ifg=0 THEN 2050 PRINT USING "16X,""Tube type : f : PLAIN""" 2070 ELSE PRINT USING "16X,""Tube type : FINNED""" PRINT_USING "16X,""Fin pitch, width, and height (mm): "",DD.D,2X.Z.DD.2X.Z 2080 2090 .DD":Fp.Fw.Fh 2100 END IF 2110 Ĵ=Ĵ IF Iov=1 THEN 2120

PRINT IF Inf=1 THEN 2130 2!40 2150 PRINT USING "10X,""Data Vu Uo Но Qρ ٧v F Nr 2160 2170 PRINT USING "10X."" # (m/s) (W/m^2-K)(W/m^2-K) (W/m`2) (m/s)""" ELSE PRINT USING ":0X.""Data Vu PRINT USING "10X."" # (m/s) 2180 IJο Ho Qp Vv""" 2190 (m/s)"" (H/m 2) (W/m^2-K) (W/m²-K) END IF Sx=0 Sy=0 Sxs=0 Sxy=0 Repeat: ! J=J+1 IF Im=1 THEN BEEP INPUT "LIKE TO CHECK NG CONCENTRATION (1=Y,0=N)?",Ng BEEP BEEP INPUT "ENTER FLOWMETER READING".Fm OUTPUT 709:"AR AF60 AL63 VR5" OUTPUT 709:"AS SA" ENTER 709:Etp OUTPUT 709:"AS SA" 2340 2350 2360 2370 2380 2390 2400 Vtran=0 FOR I=1 TO 50 ENTER 709:Vt Vtran=Vtran+Vt NEXT I 2410 Vtran=Vtran/50 DUTPUT 709:"AS SA" ENTER 709:Bvol DUTPUT 709:"AS SA" ENTER 709:Bamp ENTER 709:Bamp 2420 2430 2440 2450 2460 Iwt=0 THEN DUTPUT 709:"AR AF20 AL24 VR1" 2470 IF IF IWT=U IMEN UUIPUT 709:"AR AF20 AL24 VR1" IF IWt=1 THEN DUTPUT 709:"AR AF20 AL30 VR!" IF IWt=0 THEN Nn=4 IF IWt=1 THEN Nn=10 FOR I=0 TO Nn DUTPUT 709:"AS SA" IF I>4 THEN 2480 2490 2500 2510 2520 2530 2540 2550 2560 2570 2580 Se=0 FOR K=0 TO 10 ENTER 709;E Se=Se+E NEXT K Enf(I)=ABS(Se/10) ELSE 2590 2600 **ENTER 709:E** 2610 2620 2630 Emf(I)=ABS(E) END IF 2640 NEXT I 2650 OUTPUT 713;"T1R2E" HAIT 2 2660 ENTER 713:T11 OUTPUT 713:"T2R2E" WAIT 2 2670 2680 2690 2700 ENTER 713; T2

ADDADO SASSAS ADDADO CONTRA BARA

5

2/10 UUTPUT /13;"(182E" WAIT 2 ENTER 713:T12 2720 2730 2740 2750 2760 2770 T1 = (T11 + T12) = .5IF Ng=0 THEN 2800 BEEP INPUT "ENTER MANOMETER READING (HL.HR.HRW)".H1.Hr.Hrw 2780 Phg=H1+Hr 2790 Pwater=Hr-Hrw ELSE IF Ifg=1 OR Iut=0 THEN 2800 2810 ENTER @File; Bvol, Bamp, Vtran, Etp. Emf(0). Emf(1). Emf(2). Emf(3). Emf(4). Fm. T1. T 2820 2.Phg,Pwater 2830 END IF 2840 IF Ifg=0 AND Iwt=! THEN ENTER @File:Bvol.Bamp.Vtran.Etp.Emf(*).Fm.T1.T2.Ph g.Pwater 2850 IF IF J=1 OR J=10 OR J=20 OR J=Nrun THEN 2860 2870 Ng=1 ELSE 2880 Ng=0 2890 2900 END IF 2910 Tsteam=FNTvsv((Emf(0)+Emf(1))*.5) ! COMPUTE STEAM TEMPERATURE 2920 2930 Troom=FNTvsv(Emf(3)) IF Iwt=! THEN Tum=0. FOR I=0 TO 5 Tu(I)=FNTvsv(Emf(I+5)) 2940 2950 2960 Twm=Twm+Tw(I) NEXT I 2970 2980 2990 Tum=Tum/6 3000 END IF 3010 Tcon=FNTvsv(Emf(4)) 3020 Psat=FNPvst(Tsteam) Rohg=13529-122*(Troom-25.85)/50 Rowater=FNRhow(Troom) 3030 3040 3050 Kowater=FNKhow(!room) Ptest=(Phg=Rohg=Pwater=Rowater)=9.799/1000 Pmm=Ptest/133.322 Pkm=Ptest=1.E-3 Pks=Psat=1.E-3 Pkt=FNPvsv(Vtran)=1.E-3 Tsat=FNTvsp(Ptest) Vst=FNVvst(Tsteam) Pagest (Ptest=Page)(Ptest) 3060 3070 3080 3090 3100 3110 3120 3130 Ppng=(Ptest-Psat)/Ptest Ppst=1-Ppng 3140 3150 Mfng=1/(1+18.015/28.97*Psat/(Ptest-Psat)) Vfng=Mfng/(1.608-.608+Mfng) Mfng=Mfng+100 3160 Vfng=Vfng+100 BEEP 3170 3180 IF Iov=2 THEN 3190 3200 3210 3220 PRINT PRINT USING "10X,""Data set number - "".DD":J PRINT 3230 END IF IF IOV=2 AND Ng=1 THEN PRINT USING "10X,"" 3240 3250 G %""" Ρ Psat Ptran Toeas Tsat Ν 3260 PRINT USING "10X."" (mm) (kPa) (kPa) (kPa) $\langle C \rangle$ (C) Molal Mass 3270 PRINT USING "10X,5(3D.DD,2X).2(3D.DD.2X).2(M3D.D,2X)";Pmm,Pkm,Pks,Pkt,Tste

No. 190

SANKANA SA

am.Tsat.Vfng.Mfng 3280 PRINT 3290 3300 END IF IF Mfng>.5 THEN BEEP 3310 3320 3330 PRINT IF Im=1 THEN 3340 BEEP PRINT PRINT USING "10X.""Energize the vacuum system 3350 3360 3370 BEEP 3380 3390 INPUT "OK TO ACCEPT THIS RUN (1-Y,0=N)?",Ok IF_Ok=0 THEN BEEP 3400 DISP "NOTE: THIS DATA SET WILL BE DISCARDED!! " MAIT 5 GOTO 2280 3410 3420 3430 END IF END IF 3440 3450 3460 END IF IF Im-1 THEN IF Ifg=1 OR Iwt=0 THEN OUTPUT ƏFile;Bvol,Bamp,Vtran.Etp,Emf(0).Emf(1).Emf(2).Emf(3).Emf(4).Fm.T1. 3470 3480 3490 T2.Phg,Pwater 3500 END IF 3510 IF Ifg=0 AND Iwt=1 THEN QUTPUT @File:Bvol.Bamp,Vtran,Etp.Emf(*),Fm,T1.T2,P hg,Pwater 3520 END END IF 3530 3540 ! 3550 IF Ifg=0 AND Iwt=1 THEN OUTPUT @File1;Tw(*) ANALYSIS BEGINS TI=FNTvsv(Emf(2)) Grad=FNGrad((T1+T2)*.5) To=Ti+ABS(Etp)/(10=Grad)*1.E+6 Er1=ABS(Ti=T1) PRINTER IS 1 PRINT USING """T1 (QCT) = " PRINT USING """T1 (TC) = " 3560 3570 3580 3590 3600 = "".DD.3D";T1 3610 3620 3630 - "".DD.3D":Ti IF Er1>.5 THEN BEEP 3640 3650 PRINT "QCT AND TC DIFFER BY MORE THAN 0.5 C" BEEP 3660 3670 3680 INPUT "OK TO GO AHEAD (1=Y.0=N)?",Ok1 INFOI "OK TO GU HHEAD (T=T.U=N)/", SKT END IF PRINT USING """DT (QCT) = "".Z.3D";T2-T1 PRINT USING """DT (T-PILE) = "".Z.3D";T0-Ti IF Ok 1=0 AND Er1>.5 THEN 5100 Er2=ABS((T2-T1)-(T0-Ti))/(T2-T1) IF Er2>.05 THEN 3690 3700 3710 3720 3730 3740 3750 BEEP PRINT "QCT AND T-PILE DIFFER BY MORE THAN 5%" BEEP 3760 3770 INPUT "OK TO GO AHEAD (1-Y,0-N)?",0k2 IF 0k2-0 AND Er2>.05 THEN 5100 3780 3790 END IF PRINTER IS 701 3800 IF Itm=1 THEN 3810 T11=T1 3820 T20=T2 END IF IF Itm-2 THEN 3830 3840 3850 T1i=Ti

਼੍ਰੇ

3860 T2o=To END IF IF Itm=3 THEN T1:=(T1+T1)*.5 3870 3880 3890 T2o=(T2+To)*.5 END IF 3900 3910 Tavg=(T1i+T2o)*.5 3920 3930 3940 3950 Cpu=FNCpu(Tavg) Rhou=FNRhou(Tavg) Md=5.00049E-3+6.9861937E-3+Fm 3960 3970 Md=Md+(1.0365-1.96644E-3+Tavg+5.252E-6+Tavg^2)/.995434 Hf=Md/Rhow Vw=Hf/(PI+Di^2/4) T2o=T2o-(.0138+.001+Vw^2) Q=Md+Cpw+(T2o-T1i) Qp=Q/(PI+Do+L) 3980 3990 4000 4010 4020 Ku=FNKu(Tavg) Muw=FNMuw(Tavg) 4030 Rei=Rhow=Vw=Di/Muw 4040 4050 Prw=FNPrw(Tavg) Fet=0. 4060 Fe2=0. 4070 4080 CF=1. Lt=1. Hi=Ku=Ci/Di=Rei^.8=Pru`.3333=Cf Dt=Q/(PI=Di=(L+L1=Fe1+L2=Fe2)=Hi) Cfc=(Muw/FNMuw(Tavg+Dt))^.14 IF ABS((Cfc-Cf)/Cfc)>.01 THEN Cf=(Cf+Cfc)=.5 4090 4100 4110 4120 4130 4140 4150 GOTO 4090 END IF IF Ife=0 THEN GDTO 4250 P1=PI+(D1+Do) 4160 4170 A1=(Do-Di)*PI*(Di+Do)*.5 4180 4190 M1=(Hi+P1/(Kcu+A1))^.5 P2=PI=(Di+Dr) 4200 ; A2=(Dr-Di)*PI*(Di+Dr)*.5 M2=(Hi*P2/(Kcu*A2))^.5 Fe1=FNTanh(M1*L1)/(M1*L1) 4210 4220 4230 4240 4250 4250 4260 4270 Fe2=FNTanh(M2=L2)/(M2=L2) Lmtd=(T2o-T1i)/LOG((Tsteam-T1i)/(Tsteam-T2o)) Uo=Q/(Lmtd*PI*Do*L) Ho=!/(1/Uo-Do*L/(Di*(L+L1*Fe1+L2*Fe2)*Hi)-Rm) Dtc=Q/(PI*Di*(L+L1*Fe1+L2*Fe2)*Hi) IF ABS((Dtc-Dt)/Dtc)>.01 THEN 4090 4280 4290 4300 Hfg=FNHfg(Tsteam) Two=Tsteam-Gp/Ho 4310 4320 Tfilm=Tsteam/3+2+Two/3 Kf=FNKw(Tfilm) Rhof=FNRhow(Tfilm) 4330 4340 Huf=FNHuw(Tfilm) 4350 Hpg=.651=Kf=(Rhof^2=9.81=Hfg/(Muf=Do=Gp))^.3333 Y=Hpg=Gp^.3333 4360 4370 X=Qp 4380 4390 Sx=Sx+X Sy=Sy+Y 4400 4410 Sxs=Sxs+X^2 Sxy=Sxy+X+Y OUTPUT 0Filep:Qp.Ho Q1=500 ! TO BE MODIFIED Qloss=Ql/(100-25)+(Tsteam-Troom) ! TO BE MODIFIED 4420 4430 4440 4450 4460 Hfc=FNHf(Tcon)

e. .

NACCOLAR.

4470 Hf=FNHf(Tsteam) 4480 Mdv=0 4490! Bp=(Bvol+100)`2/5.75 ! BOILER POWER IN Watts 4500 Bp=(Bvol+100)^2/5.76 4510 Mdvc=((Bp-Qloss)-Mdv+(Hf-Hfc))/Hfg 4520 IF ABS((Mdv-Mdvc)/Mdvc)>.01 THEN Mdv=(Mdv+Mdvc)+.5 4530 4540 4550 GOTO 4510 END IF 4560 Mdv=(Mdv+Mdvc)=.5 4570 Vg=FNVvst(Tsteam) Vv=Mdv=Vg/Ax IF Inf=1 THEN 4580 4590 F=(9.799+Do+Muf+Hfg)/(Vv^2+Kf+(Tsteam-Two)) 4600 4610 Nu=Ho#Do/Kf Ret=Vv*Rhof*Do/Muf 4620 Nr=Nu/Ret[^].5 4630 4640 END IF IF Inf-1 THEN OUTPUT ONrf;F.Nr IF Inf-1 THEN OUTPUT ONrf;F.Nr IF Iov-2 THEN PRINT USING "10X."" T (Inlet) Delta-T""" PRINT USING "10X."" QCT TC QCT T-PILE""" PRINT USING "10X.2(DD.DD.2X).2(Z.3D.2X)";T1,Ti,T2-T1,To-Ti PRINT USING "10X,"" Vw Rei Hi Uo 4650 4560 4670 4680 4690 Ho 4700 Vv" PRINT USING "10X.Z.DD.1X.5(MZ.3DE.1X).MZ.DD"; Vw.Rei,Hi,Uo,Ho,Qp,Vv 4710 END IF IF Iov=1 THEN IF Inf=1 THEN 4720 4730 4740 PRINT USING "11X.DD.2X,Z.DD.2X,2(5D.D.2X).Z.3DE.1X.Z.DD.2(1X.3D.DD)":J.Yu. 4750 Uo.Ho.Qp.Vv.F.Nr 4760 ELSE 4770 PRINT USING "11X.DD.2X.Z.DD.2X.2(MD.4DE.2X),Z.3DE.3X,Z.DD"; J.Vw.Uo.Ho.Gp.V 4780 END IF 4790 4800 IF Im=1 THEN BEEP 4810 INPUT "WILL THERE BE ANOTHER RUN (1=Y,0=N)?".Go_on 4820 4830 Nrun=J IF Go_on=1 THEN Repeat 4840 ELSE IF J<Nrun THEN Repeat 4850 4860 END IF IF Ifg=0 THEN 4870 4880 PRINT 4890 4900 S1=(Nrun+Sxy-Sy+Sx)/(Nrun+Sxs-Sx`2) Ac=(Sy-SI=Sx)/Nrun PRINT USING "10X.""Least-Squares Line for Hnu vs q curve:" PRINT USING "10X."" Slope = "",MD.4DE";Sl PRINT USING "10X."" Intercept = "",MD.4DE";Ac 4910 4920 4930 4940 4950 END IF IF Im=1 THEN BEEP 4960 4970 PRINT 4980 PRINT USING "10X.""NOTE: "".ZZ."" data runs were stored in file "".10A":J. 4990 D_fileS 5000 E END IF BEEP 5010 PRINT 5020 5030 PRINT USING "10X.""NOTE: "".ZZ."" X-Y pairs were stored in plot data file

1924

CARE READ

Same Road and

うちいいつもう

"".10A":J.P_file\$ 5040 IF Inf=1 THEN PRINT USING "16X.ZZ."" pairs of Nr-F are stored in file "".14A": J.Nrfs 5050 END IF ASSIGN File TO * ASSIGN File1 TO * ASSIGN Filep TO * 5060 5070 5080 5090 5100 END 5110 5120 DEF FNPvst(Tsteam) DIM K(8) DATA -7.691234564,-26.08023696,-168.1706546,64.23285504.-118.9646225 5130 5140 DATA 4.16711732,20.9750676,1.E9.6 5150 READ K(+) 5160 5170 T=(Tsteam+273.15)/647.3 Sum=0 FOR N=0 TO 4 5180 Sum=Sum+K(N)+(1-T)^(N+1) NEXT_N 5190 5200 5210 Br=Sum/(T+(1+K(5)+(1-T)+K(6)+(1-T)*2))-(1-T)/(K(7)+(1-T)*2+K(8)) 5220 5230 5240 Pr=EXP(Br) P=22120000=Pr RETURN P 5250 5260 5270 FNEND DEF FNHfg(T) Hfg=2477200-2450*(T-10) RETURN Hfg 5280 5290 5300 FNEND DEF FNMuw(T) A=247.8/(T+133.15) Hu=2.4E-5+10^A 5310 5320 RETURN Mu 5330 FNEND DEF FNVvst(Tt) P=FNPvst(Tt) T=Tt+273.15 5340 5350 5360 5370 5380 X=1500/T F1=1/(1+T+1.E-4) 5390 F2=(1-EXP(-X))^2.5=EXP(X)/X^.5 B=.0015=F1-.000942=F2-.0004882=X K=2=P/(461.52=T) V=(1+(1+2=B=K)^.5)/K 5400 5410 5420 5430 5440 RETURN V 5450 FNEND DEF FNCpw(T) Cpw=4.21120858-T*(2.26826E-3-T*(4.42361E-5+2.71428E-7*T)) RETURN Cpw+1000 5460 5470 5480 FNEND 5490 DEF FNRhow(T) 5500 5510 5520 Ro-999.52946+T+(.01269-T+(5.482513E-3-T+1.234147E-5)) RETURN Ro 5530 FNEND 5540 DEF FNPru(T) 5550! 5560 Prw=10^(1.09976605-T*(1.3749326E-2-T*(3.968875E-5-3.45026E-7*T))) Pru-FNCpu(T)+FNMuu(T)/FNKu(T) 5570 RETURN Pru 5580 FNEND DEF FNKw(T) Kw=.5625894+T+(2.2964546E-3-T+(1.509766E-5-4.0581652E-8+T)) X=(T+273.15)/273.15 Kw=-.92247+X+(2.8395-X+(1.8007-X+(.52577-.07344+X))) 5590 56001 5610 5620 RETURN KW 5630

「「「ない」」「「「「「「」」」」」

No Contraction

Contraction of the second s

5640 5650 5660 FNEND DEF FNTanh(X) P=EXP(X) Q=EXP(-X) 5670 5680 5690 5700 Tanh=(P+Q)/(P-Q) RETURN Tanh FNEND 5710 5720 5730 5740 5750 DEF FNTvsv(V) COM /Cc/ C(7) Sum=C(0) FOR I=1 TO 7 Sum=Sum+C(I)+V^I NEXT I T=V*(.02635206856-V*(9.7351313E-7-V*6.576805E-11)) RETURN Sum 5760 5770! 5780 5790 5800 RFNEND DEF FNHf(T) Hf=T*(4.203849-T*(5.88132E-4-T*4.55160317E-6)) RETURN Hf*1000 5810 5820 5830 5840 5850 FNEND DEF FNGrad(T) Grad=37.9853+.104388*T RETURN Grad 5860 5870 FNEND DEF FNTvsp(P) Tu-110 5880 5890 5900 5910 TI=10 Ta=(Tu+T1)*.5Ta=(Tu+T)=.5 Pc=FNPvst(Ta) IF ABS((P-Pc)/P)>.001 THEN IF Pc<P THEN TI=Ta IF Pc>P THEN Tu=Ta GOTO 5910 END IF PETUPN Ta 5920 5930 5940 5950 5950 5960 5970 RETURN Ta 5980 5990 FNEND 6000 6010 6020 6030 DEF FNPvsv(V) P=8133.5133+2.236051E+4*V RETURN P FNEND

a ha ha h

A CONTRACTOR OF A

ALL DAY DAY DAY

AND A THE PROPERTY INSURANCE IN THE PROPERTY IN

N N L . I

PROGRAM TCAL

100 ! FILE NAME: TCAL 110 ! REVISED: December 11, 1983 120 ! COM /Ce/ C(7) 130 DIM Emf(10).T(10).D(10) DATA 0.10086091.25727.94369.-767345.8295.78025595.81 140 150 160 DATA -9247486589,6.97688E11.-2.66192E13.3.94078E14 170 180 READ C(+) CLEAR 709 190 BEEP 200 210 220 230 240 250 260 270 INPUT "ENTER MONTH, DATE AND TIME (MM:DD:HH:MM:SS)", BS J=0 OUTPUT 709:"TD":BS OUTPUT 709:"TD" ENTER 709:AS PRINT USING "10X.""Month. date and time = "",14A";AS BEEP INPUT "ENTER INPUT MODE (1=3054A, 2=FILE)", Im 280 290 300 IF Im=1 THEN BEEP INPUT "GIVE A NAME FOR DATA FILE".D_files 310 320 CREATE BDAT D_file\$.5 ELSE 330 BEEP 340 350 INPUT "GIVE NAME OF EXISTING FILE", D_files BEEP 360 370 380 INPUT "ENTER NUMBER OF DATA RUNS STORED", Nrun END IF BEEP INPUT "GIVE A NAME FOR PLOT FILE".P_file\$ CREATE BDAT P_file\$.5 ASSIGN @Plot TO P_file\$ ASSIGN @File TO D_file\$ 390 400 410 420 IF Im=1 THEN BEEP 430 440 BEEP INPUT "ENTER BATH TEMPERATURE".T_bath OUTPUT 709;"AR AF20 AL30" FOR I=0 TO 10 OUTPUT 709:"AS SA" ENTER 709:Emf(I) NEXT I OUTPUT 712."TIP2E" 450 460 470 480 490 500 OUTPUT 713;"T1R2E" 510 520 530 540 550 HAIT 2 ENTER 713:T1 OUTPUT 713:"T2R2E" HAIT 2 560 ENTER 713:T2 OUTPUT PFile:T_bath.Emf(*).T1.T2 570 580 ELSE 590 ENTER_9File:T_bath.Enf(+).T1.T2 END IF 600 510 620 630 J=J+1 Dwa=0 FOR I=0 TO 10 T(I)=FNTvsv(ABS(Emf(I))) 640 650 660 570 D(I)=T_bath-T(I) IF I>4 THEN Dwa=Dwa+D(I) NEXT_I Dwa=Dwa/6 680

Dsa=(D(0)+D(1))+.5 690 700 710 OUTPUT @Plot;T_bath.Dsa.Dwa PRINT PRINT PRINT USING "10X.""Data set number = "".DD";J PRINT USING "10X.""Bath T (C) QCT-1 (C) QCT-2 PRINT USING "10X.3(3D.3D.7X)":T_bath.T1,T2 PRINT USING "10X."Thermocouple readings (Deg C):""" PRINT USING "10X.6(3D.DD.3X).16X":T(*) PRINT USING "10X.""Discrepancies (Deg C):""" PRINT USING "11X.6(MZ.DD.4X).15X":D(*) PCEP 720 730 QCT-2 (C)""" 740 750 760 770 780 790 800 BEEP IF In-1 THEN INPUT "ARE YOU TAKING MORE DATA (1-Y.O-N)?".Go_on IF Go_on-1 THEN 430 810 820 830 ËLSË 840 850 IF J<Nrun THEN 430 END IF 860 870 PRINT IF Im=1 THEN PRINT USING "10X.""NOTE: "".DD,"" data sets are stored in file "".14A":J.D 880 _file\$ 390 ELSE 900 PRINT USING "10X.""NOTE: Above analysis was performed from file "".14A":D_ file\$ END IF PRINT USING "16X.""Plot data are stored in file "".10A":P_fileS ASSIGN @File 10 * ASSIGN @Plot 10 *-910 920 930 940 950 END 960 970 DEF FNTvsv(Emf) COM /Cc/ C(7) Sum=C(0) FOR I=1 TO 7 980 990 Sum=Sum+C(I)=Enf^I NEXT I 1000 1010 RETURN Sum 1020 1030 FNEND

LIST OF REFERENCES

15125455501 +222272231

ļ

1

1.	Bergles. R.E., and Jensen, M.K., "Enhanced Single-phase Heat Transfer for Ocean Thermal Energy Conversion Systems," HTL-13, Iowa State University, Ames, Iowa, April, 1977.
2.	Marto, P.J., "Heat Transfer and Two-phase Flow During Shell-side Condensation," <u>ASME•JSME</u> <u>Thermai</u> <u>Engineering Joint Conference Proceedings</u> , Honolulu, Vol. 2, 1983, pp. 561-591.
3.	Reilly, D.J., <u>An Experimental Investigation of Enhanced Heat Transfer on Horizontal Condenser Tubes</u> , AS Thesis, Naval Postgraduate School, Monterey, California, March, 1978.
4.	Kanakis, G.D., <u>The Effect of Condensate Inundation on</u> <u>Steam Condensation Heat Transfer of Alle-Wrapped</u> <u>Tubing</u> , <u>MS</u> Thesis, Naval Postgraduate School, Monterey, California, June, 1983.
5.	Krohn. R.L., <u>An Experimental Apparatus to Study</u> Enhanced <u>Condensation Heat Transfer</u> of Steam on <u>Horizontal Tubes</u> , <u>Ms Thesis</u> , <u>Naval</u> Postgraduate School, Monterey, California, June, 1982.
6.	Yau, K.K., and Rose, J.W., <u>Effects of Fin Spacing and</u> Drainage Strip on the <u>Condensation</u> Heat-transfer <u>Performance</u> of <u>Horizontal</u> Low Integral-fin Tubes, Picject Report, Department of Mechanical Engineering, Queen Mary College, University of London, 1982
7.	Graber, K. A., <u>Condensation Heat Transfer of Steam on a</u> <u>Single Horizontal Tube</u> , MS Thesis, Naval Postgraduate School, Monterey, California, June, 1983.
8.	Nobbs, D.W., <u>The Effect of Downward Vapour Velocity</u> and <u>Inundation</u> on the <u>Condensation</u> Rates of <u>Horizontal</u> <u>Tubes</u> and <u>Tube</u> Banks, Ph.D Dissertation, University of Bristol, Bristol, England, April, 1975.
9.	Barrington, A.E., <u>High Vacuum Engineering</u> , Prentice-Hall, 1963, p. 182.
10.	Fujii, T., Honda, H., and Oda, K., "Condensation of Steam On a Horizontal Tube," <u>Condensation Heat</u> <u>Transfer</u> , ASME New York, 1979, pp. 35-43.
11.	Incropera, F.F., and Dewitt, D.P., <u>Fundamentals of</u> <u>Heat Transfer</u> , Wiley, 1981, pp. 406-407.

12. Thomas, D.G., "Condenser Tube," U.S. Patent 3,358,750, United States Patent Office, 1967. 13. Gebhart, B., <u>Heat Transfer</u>, 2nd ed., McGraw-Hill, 1971, p.35.

INITIAL DISTRIBUTION LIST

I STATES

12-22-22-21

18-20-50 F

τ,

		No.	Copies
1.	Defense Technical Information Center Cameron Station Alexandria VA 22314		2
2.	Library, Code 0142 Naval Postgraduate School Monterey CA 93943		2
3.	Department Chairman, Code 69 Department of Mechanical Engineering Naval Postgraduate School Monterey CA 93943		1
4.	Professor P. J. Marto, Code 691x Department of Mechanical Engineering Naval Postgraduate School Monterey CA 93943		5
5.	Dr. John W. Rose Department of Mechanical Engineering Queen Mary College University of London London E1 4NS England		1
6.	Dr. Win Aung Program Director for Heat Transfer Division of Engineering National Science Foundation Washington DC 20550		1
7.	Dr. A. S. Wanniarachchi, Code 59Wa Department of Mechanical Engineering Naval Postgraduate School Honterey CA 93943		1
8.	Mr. R. W. Kornbau Code 2721 David Taylor Naval Ship Research and Development Center Bethesda ND 20084		2
9.	LCDR Raymond L. Krohn, USN Portsmouth Naval Shipyard Portsmouth, N. H. 03801		1
10.	LT Kenneth A. Graber, USN Naval Education and Training Center Newport RI 02840		1

11. IT William M. Poole, USN U.S. Naval Ship Repair Facility Box 34 FPO San Francisco CA 96651

2002222002

2000(2001 2002022) V200000-1

Ĵ

 ۰.









Û,

