

AD-A128-203

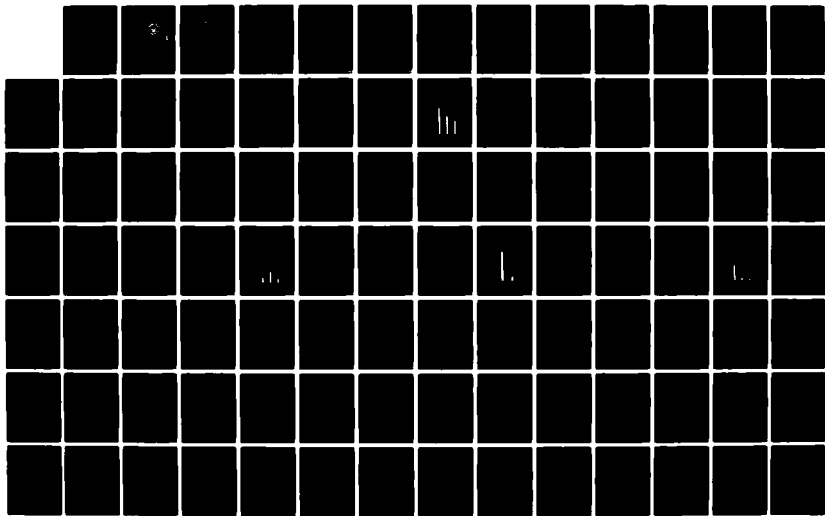
CORRECTIONS AND IMPROVEMENTS TO THE INTERACTIVE  
COMPUTER PROGRAM FOR THE (U) NAVAL POSTGRADUATE SCHOOL  
MONTEREY CA R M HILL MAR 83

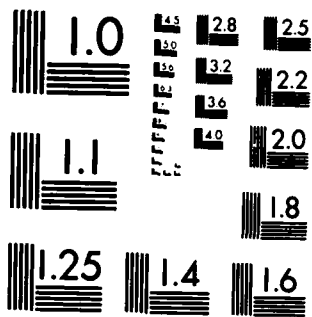
1/2

UNCLASSIFIED

F/G 9/2

NL





MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS 1963-A

2

# NAVAL POSTGRADUATE SCHOOL

Monterey, California



ADA 128203

DTIC  
S ELECTE  
MAY 17 1983  
D

## THESIS

CORRECTIONS AND IMPROVEMENTS TO THE  
INTERACTIVE COMPUTER PROGRAM FOR THE SURVIVABILITY  
EVALUATION OF AIRCRAFT CONCEPTUAL DESIGNS ( VISAP )

by

Ronald Maxwell Hill

March 1983

Thesis Advisor:

R. E. Ball

Approved for public release, distribution unlimited.

FILE COPY

83 05 16 123

## **DISCLAIMER NOTICE**

**THIS DOCUMENT IS BEST QUALITY PRACTICABLE. THE COPY FURNISHED TO DTIC CONTAINED A SIGNIFICANT NUMBER OF PAGES WHICH DO NOT REPRODUCE LEGIBLY.**

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER	2. GOVT ACCESSION NO. <b>A128203</b>	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) Corrections and Improvements to the Interactive Computer Program for the Survivability Evaluation of Aircraft Conceptual Designs (VISAP)		5. TYPE OF REPORT & PERIOD COVERED Master's Thesis March 1983
7. AUTHOR(s) Ronald Maxwell Hill		6. PERFORMING ORG. REPORT NUMBER
9. PERFORMING ORGANIZATION NAME AND ADDRESS Naval Postgraduate School Monterey, California 93940		8. CONTRACT OR GRANT NUMBER(s)
11. CONTROLLING OFFICE NAME AND ADDRESS Naval Postgraduate School Monterey, California 93940		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		12. REPORT DATE March 1983
		13. NUMBER OF PAGES 182
16. DISTRIBUTION STATEMENT (of this Report)  Approved for public release, distribution unlimited.		15. SECURITY CLASS. (of this report)
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Aircraft Non-nuclear Survivability Computer Conceptual Design Evaluation		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) A computer program for assessing the survivability of fixed wing aircraft in the conceptual design phase was developed at the Naval Postgraduate School by Ball and Hesser in 1982. The program was called VISAP (Vought Interactive Survivability Assessment Program). This thesis presents corrections and improvements made to VISAP by the author. These corrections and improvements include improved efficiency and friendliness of the program from the user's viewpoint, enhanced output, and the incorporation of graphics to aid in the assessment and evaluation of aircraft conceptual design.		

DD FORM 1473

1 JAN 73

EDITION OF 1 NOV 65 IS OBSOLETE

S/N 0102-014-6001

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

Approved for public release, distribution unlimited

Corrections and Improvements to the  
Interactive Computer Program for the Survivability  
Evaluation of Aircraft Conceptual Designs ( VISAP )

by

Ronald Maxwell Hill  
Lieutenant Commander, United States Navy  
B.S., United States Naval Academy, 1970  
M.S.B.A., University of Northern Colorado, 1978

Submitted in partial fulfillment of the requirements for  
the degree of

MASTER OF SCIENCE IN AERONAUTICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL  
March 1983

Author:

R M Hill

Approved by:

N E Bell

Thesis Advisor

Donald M. Lester

Department of Aeronautics

A. Dyer

Dean of Science and Engineering

ABSTRACT

A computer program for assessing the survivability of fixed wing aircraft in the conceptual design phase was developed at the Naval Post-graduate School by Ball and Hesser in 1982. The program was called VISAP (Vought Interactive Survivability Assessment Program). This thesis presents corrections and improvements made to VISAP by the author. These corrections and improvements include improved efficiency and friendliness of the program from the user's viewpoint, enhanced output, and the incorporation of graphics to aid in the assessment and evaluation of aircraft conceptual design.

<b>Accession For</b>	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By _____	
Distribution/	
Availability Codes	
Dist	Avail and/or Special
A	



## TABLE OF CONTENTS

I. INTRODUCTION -----	7
II. NATURE OF THE PROBLEM -----	11
III. SOLUTIONS -----	14
APPENDIX A. VARIABLES, SUBROUTINES AND DEFINITIONS -----	21
APPENDIX B. SAMPLE INSTRUCTION MANUAL AND ASSESSMENTS -----	35
APPENDIX C. FLOW CHARTS -----	55
APPENDIX D. VISAP AND DISVIS EXEC PROGRAM LISTINGS -----	57
APPENDIX E. ESCORT AND ESCPLT PROGRAM LISTINGS -----	58
APPENDIX F. STRIKE AND STRPLT PROGRAM LISTINGS -----	102
APPENDIX G. SUPPORT AND SUPPLT PROGRAM LISTINGS -----	140
LIST OF REFERENCES -----	180
INITIAL DISTRIBUTION LIST -----	181



## LIST OF FIGURES

1. Example Plot -----	18
2. Sample Escort Plot -----	43
3. Sample Escort Baseline Output -----	44
4. Sample Escort 1st Design Output -----	45
5. Sample Escort 2nd Design Output -----	46
6. Sample Strike Plot -----	47
7. Sample Strike Baseline Output -----	48
8. Sample Strike 1st Design Output -----	49
9. Sample Strike 2nd Design Output -----	50
10. Sample Support Plot -----	51
11. Sample Support Baseline Output -----	52
12. Sample Support 1st Design Output -----	53
13. Sample Support 2nd Design Output -----	54
14. External Program Flow Chart -----	55
15. Internal Program Flow Chart -----	56

## ACKNOWLEDGMENT

To Professor Robert E. Ball for his encouragement and direction which always kept the objective in sight.

To my bride of ten months, Mary Ann, for her love, devotion, and limitless patience. Her advice was timely, consistent, and unfailingly wise.

## I. INTRODUCTION

### A. SURVIVABILITY ( P(s) )

Aircraft combat survivability is defined as the capability of an aircraft to avoid and/or withstand a man-made hostile environment. This ability to avoid or withstand the hostile environment is a function of several factors both inherent in and external to the aircraft. Survivability is quantifiable using basic probability theory and can be expressed as unity minus the product of the aircraft's susceptibility and the aircraft's vulnerability.

$$P(s) = 1 - \text{Susceptibility} \times \text{Vulnerability} \quad (1)$$

#### 1. Susceptibility ( P(h) )

Susceptibility is the aircraft's inability to avoid the hostile environment. It can be expressed as the probability that the aircraft is hit (P(h)), it is influenced by a multitude of factors. Generally, these factors consist of the threat activity, the threat sensors, the threat tracking ability and the threat propagator. These factors can also be quantified. They can be expressed as the probability of activity (P(a)), the probability of detection (P(d)), the probability of conversion (P(c)), and the probability of damage (P(dam)).

$$P(h) = P(a) \times P(d) \times P(c) \times P(\text{dam}) \quad (2)$$

Susceptibility can be reduced by one or more means. Prominent are threat suppression (reduction of the threat's activity or ability to act), signature reduction (minimizing the aircraft's visual, aural, and electromagnetic emissions or reflections), and overt countermeasures (interference with the threat's ability to track or engage the aircraft).

## 2. Vulnerability ( P(k/h) )

Once hit by a damage causing mechanism, such as a fragment or projectile, the reaction of the aircraft is dependent upon its vulnerability. The vulnerability levels may range from no effect through catastrophic destruction, with intermediate effects including but not limited to mission degradation, system or subsystem malfunctions, and component failures.

Vulnerability is often measured using the concept of vulnerable area. An aircraft presents a projected area (A(p)) depending on the aspect of the observer or tracking system. Each aircraft critical component has its own vulnerable area that contributes to the total aircraft vulnerable area (A(v)). The vulnerability of the aircraft can also be measured by the ratio of the aircraft vulnerable area to the aircraft presented area.

$$P(k/h) = A(v) / A(p) \quad (3)$$

P(k/h) is the probability the aircraft is killed given a hit.

### B. VISAP

"The Development of an Interactive Computer Program for the Evaluation of Aircraft Conceptual Designs" [Ref. 1] was the result of the compilation of numerous efforts to perform survivability assessments in the conceptual design phase utilizing deterministic models. The computer programs, collectively called the VISAP (Vought Interactive Survivability Assessment Program) program, eloquently allow the designer or analyst to investigate the effects on survivability of altering, singly or in groups, the aircraft's design features, its vulnerability and susceptibility reduction features, and/or the threat environment parameters. Results of a single sortie and

a campaign analysis and the incremental increases to the aircraft's take-off gross weight are used as measures of effectiveness.

VISAP is also the filename of the CMS control EXEC designed for use on the Naval Postgraduate School's IBM 3033 computer. When executed, it presents the user with the choice of one of three aircraft types to analyze. These types are Fighter Escort, Long Range Strike, and Close Air Support. Each type is assessed by independent programs with filenames of ESCORT, STRIKE, and SUPPORT respectively.

Each program solves the survivability equation using values calculated from the design parameters chosen by the user from "menus" incorporated into the programs and automatically displayed on the user's terminal at the appropriate time during program execution. Subsidiary routines and subroutines either correlate the inputs with tabulated data or do deterministic calculations to produce values for, ultimately, the probability of survival  $P(s)$  for a design of an aircraft type against predetermined threats. Three subroutines are utilized to determine the results of the single sortie of an aircraft, to conduct a campaign analysis consisting of several flights by many aircraft, and to show a comparison between the new and the original gross weights.

Parameter values are displayed on the terminal while the user is running the program. Changes made are immediately indicated, and the values calculated from the changes are also displayed when appropriate. In addition, upon completion of a design, a hard copy printout may be obtained if desired. This printout contains the susceptibility and vulnerability reduction features, values for  $P(s)$ ,  $P(d)$ ,  $P(h)$ , and  $P(k/h)$ ,

results of the campaign analysis, the baseline takeoff gross weight, and the enhanced gross weight.

#### C. CONCLUSION

VISAP was an immense improvement over the previous requirements to correlate masses of empirical and analytic data. The elimination of time consuming, tedious, and, therefore, error prone hand calculations is, of course, the principle benefit of the programs.

## II. NATURE OF THE PROBLEM

### A. INTRODUCTION

VISAP was used at the Naval Postgraduate School in course AE-3251, Aircraft Combat Survivability during the Spring Quarter 1982. Students were assigned the task of analyzing survivability enhancements on the three available aircraft types. While the results of this project were generally favorable, several inadequacies were discovered in the programs. Furthermore, solicited comments from industry and government activities studying the program pointed out other errors and several suggestions for improvement. The gist of the significant errors, inadequacies, and recommendations are:

1. Erroneous output in some cases
2. Inaccuracies in the "HELP" menus
3. Excessive time to work through a design
4. Inability to save design changes from one run to the next
5. Necessity to reenter each point in the program to duplicate a design
6. Requirement to rerun an entire program to assess the effects of a change to a parameter
7. Limited data on printouts making comparisons between the design and effects difficult
8. Questionable validity of the results
9. No provisions for cost information provided
10. Lack of graphical presentation of results

## B. SPECIFICS OF THE PROBLEM

Difficulties with VISAP in general are categorized as follows:

1. The data output and validity of results are suspect due to random and obviously erroneous results. Several minor corrections in the sub-routine programming were identified. The corrections to this and other problems will be discussed in more detail in the next chapter. The methodology used to develop the algorithms for the programs' subroutines are not questioned.

2. Inaccuracies and garbled text in the "HELP" menus were identified. Proper interpretations were researched in "The Fundamentals of Aircraft Combat Survivability Analysis and Design" [Ref. 2]. Specifically, the help menu 6's equation for  $P(s)$  was incorrect,  $P(S) = P(D)*P(H)*P(K/H)$  instead of the correct,  $P(S) = 1 - ( P(D)*P(H)*P(K/H) )$ . Also, HELP menu 3 contained a nonsense line reading, "of study as the A/C type defined them."

3. Students universally complained about an excessive amount of time to complete an evaluation. The inability to save the results of a design effort by means other than reaccessing VISAP at the beginning and having to reenter all previously chosen data was also of concern. The need for a data saving and retrieval routine, in addition to the established capability to automatically reenter the program at the completion of a run, was established. Furthermore, once the user familiarizes himself/herself with program operation, stepping through each sequence becomes redundant. Therefore, a means to automatically assess individual design changes was required.



4. Accompanying item 3 above was the necessity to expand the output. To help identify a design analysis and to correlate which parameter affected which measure of effectiveness, the printouts required design and performance information in addition to the susceptibility and vulnerability reduction features already presented.

5. To enhance industry use, cost information was recommended for inclusion in VISAP. While costing was a major emphasis in the preliminary research, it was not incorporated in Reference 1 and is also considered beyond the scope of the current project.

6. A graphical presentation of an assessment seemed a logical application of VISAP. In fact, a bar chart depicting aircraft loss rate or  $P(k)$  versus the threat types was a requirement for the AE-3251 project. A means to utilize some of the graphics utilities available at the Naval Postgraduate School was, therefore, made a requirement.

#### C. CONCLUSION

Chapter three will delve into the details of the changes and corrections made to the version of VISAP described in Reference 1. The intention of continuing work on VISAP was to improve the efficiency of the program, extend its applicability, and broaden the range of useful information produced. The basic methodologies, approach to the solution, and programming techniques were all considered suitable and, therefore, the corrections and additions are principally enhancements to the basic programs.

### III. SOLUTIONS

#### A. GENERAL

The solutions will be discussed in the same order as the problems to which they relate were delineated in Chapter II. Additionally, appendixes E-G, the program listings, have been annotated with a numbered comment card ( c ## ---- ) preceding each section that has been altered from the original version of VISAP. The number (##) in the comment refers to the like numbered statements of the following paragraphs.

#### B. SOLUTION SPECIFICS

Corrective action for the problems were developed as follows:

1. Random, erroneous output values were the result of computational errors, programming errors, and the use of mixed mode arithmetic (i.e., integer instead of real data). These errors occurred in the SORT and CAMP subroutines of all three programs. Mixed mode was also discovered in SUPPORT in the Menu 41 section on Vulnerable Area/P(k) vs. AAA, in the SRPDSM, the SRVAAA, and the SRPHR subroutines, in the STRIKE subroutines SSRPDS and ESRWT, and the ESCORT ESRVAVG and ESRWT subroutines. The affected sections and subroutines were analyzed, corrected, and now check against hand calculated values for sample cases.

2. Inaccuracies and garbled text in the "HELP" menus as mentioned in Chapter II were identified. The text with corrections has been retyped maintaining the existing format.

3. Incorporation of routines to save data and modifications of the program flow to expedite the time required to perform an evaluation have been made. Data is now retained in a disk file and is continually updated as particular parameters or values change during program execution. At program termination, or any time MENU 7, the assessment routine, is executed, current data is "dumped" to the data file. Separate files, named ESCORT DATA, STRIKE DATA, and SUPPORT DATA, are maintained for the respective aircraft types. When reentering a program, the user is given the option of using either his previously defined data or the default values specified in the declaration section of the program.

The programs are now written to cause an automatic assessment any time a variable is changed. This is accomplished with "GO TO" statements in the menus Main, 2, 3, 4, 5, and 6 which force the program to execute Menu 7, to evaluate gross weight changes (subroutines ESPWT or SSRWT), and to record all values in the data file.

Following the evaluation, when the user exits the program, the current assessment is displayed on the terminal. He/she may opt to have this information printed, then exit; reenter the program; or exit without a printout.

4. The printouts themselves include new sections. The full title of the aircraft type is spelled out. For example, "Long Range Strike Aircraft" replaces the abbreviated "Strike Aircraft" used previously. Performance features, mission parameters, and threat parameters are enumerated, in addition to the existing susceptibility and vulnerability reduction features. These additions facilitate the identification of

the cause and effect relationships between the independent design variables and the resulting changes in the survivability assessment.

The augmented printouts are produced by rewritten statements in the Exit routines' "WRITE" statements and their associated "FORMAT" statements. Furthermore, this output is identical to that displayed on the terminal screen which was discussed in objective 3. This is accomplished by incorporating repetitive "WRITE" statements with the unit codes changed to direct output to the terminal instead of the printer.

5. Graphics capability posed many possibilities and a multitude of alternatives. First, consideration had to be given to what information was to be presented. Since the Probability of Survival ( $P(s)$ ) or the Probability of Kill ( $P(k)$ ) provides a comprehensive, quantifiable evaluation of a design, the choices were immediately limited to one of these. Of the two, Probability of Kill, against each of the threat types, was arbitrarily picked since it was anticipated to show a decreasing trend for each successive design which seems more esthetically pleasing. Second, a decision concerning the format of the graph was needed. A bar chart was picked for its simplicity and to remain consistent with the AE-3251 project objectives. Third, several plotting devices are available that can be accessed either directly from VISAP or separately by the user. The dual screen IBM 3277/Tektronix 618 system at NPS was chosen due to its availability and its ability to produce both a CRT display and a hard copy printout. The user must decide upon which assessments to have plotted, and then, subsequent to exiting the program but at his/her convenience, he/she may obtain graphs of the chosen designs. Finally, the numbers of

assessments to be depicted had to be determined. In keeping with prior requirements, and in an effort to supply adequate information and yet prevent the charts from becoming cluttered, a total of three design alternatives are presented. These are indicated on the graphs by separate bars corresponding to a Baseline, a 1st Design, and a 2nd Design. Three bars corresponding to the three alternative designs are clustered vertically above the appropriate threat type. Figure 1 shows a typical plot.

VISAP MENU 8 was written to calculate the Probability of Kill against each threat type.

$$P(k) = 1. - P(s) \quad (4)$$

Menu 8, additionally incorporates routines to query the user about his plotting intentions, to provide him/her with further plotting procedure information, and to file the data required for the plots. When Menu 8 is executed, the user is informed as to how many designs he has selected for plotting (i.e., 0 of 3, 1 of 3, or 2 of 3) and is given the opportunity to access a HELP MENU 8 which was written to provide further information concerning plot procedures. If the user decides to have the current design depicted, VISAP files the plotting data in disk files named ESCPLT DATA, STRPLT DATA, or SUPPLT DATA, respectively, from the ESCORT, STRIKE or SUPPORT programs. These plot data files are distinct from the aforementioned "save" data files.

The DISSPLA (a Proprietary Software Product of Integrated Software Systems Corporation) system was utilized to write separate Fortran IV programs for each aircraft type. Named ESCPLT, STRPLT, and SUPPLT, they peruse their respective data files, format the presentation, and direct

# EXAMPLE AIRCRAFT

## Loss Rate VS. Threat Type

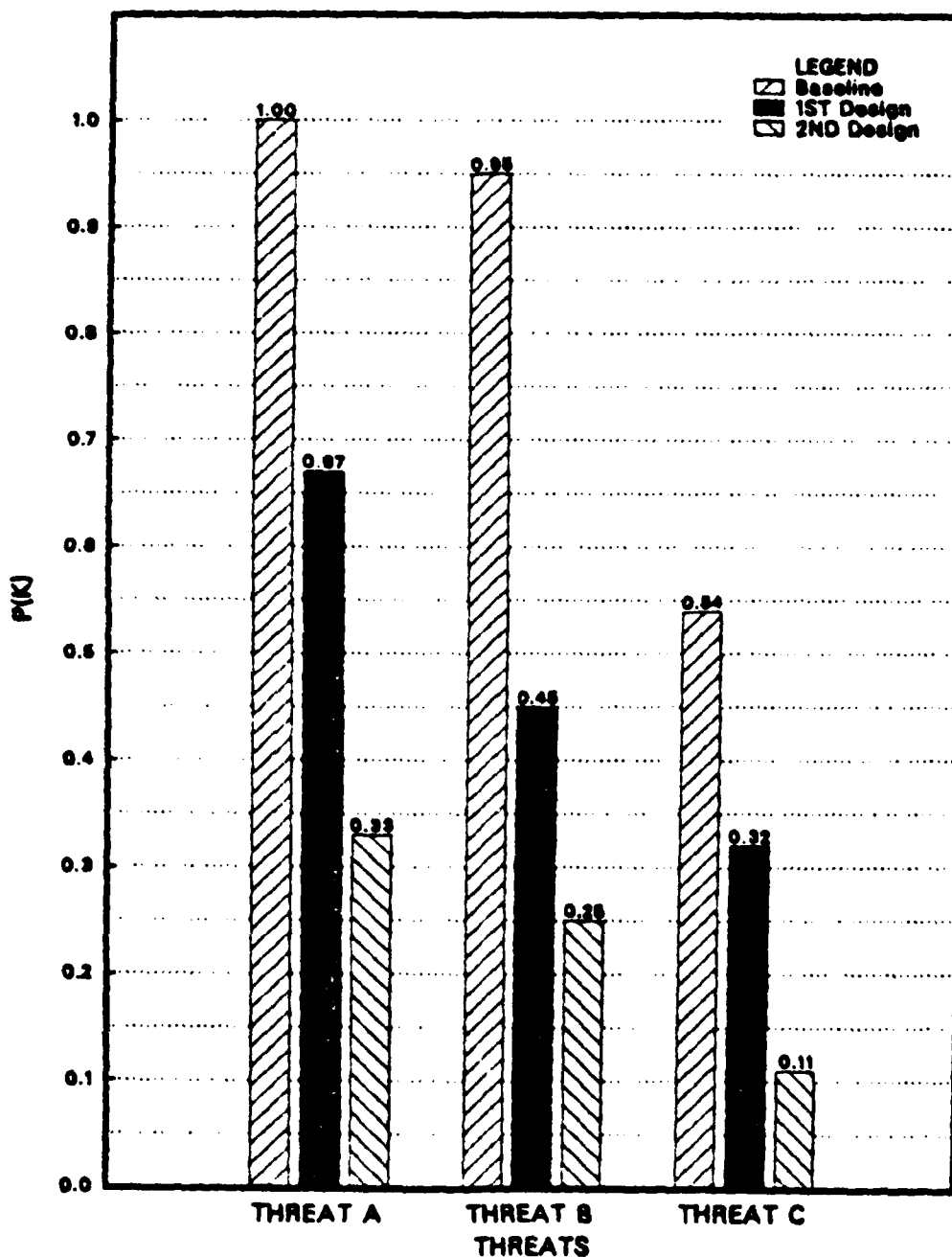


Figure 1. EXAMPLE PLOT

the output to the Tektronix 618 display screen. The user simply presses the "HARD COPY" key below the screen to obtain a printed copy assuming, of course, the screen in use is attached to a printer.

ESCPLT, etc., are controlled by a CMS EXEC called DISVIS. The user operates DISVIS in a manner similar to VISAP. When seated before a dual screen terminal, and having accessed DISVIS, he/she enters the plot desired (ESCPLT, STRPLT, or SUPPLT) and the proper graph will automatically appear on the adjoining screen. One caution must be noted. The user must have made three design selections during one terminal session prior to attempting a plot request. If fewer than three data points are filed, DISSPLA will inform the viewer that "the end of data file" on the appropriate unit number (disk) has been reached. While some sort of graph may be presented, there will probably be depicted zero values, erroneous  $P(k)$ 's or other even more erratic or undesirable output.

#### C. CONCLUSION

VISAP as currently configured is an extensive, highly versatile, and efficient computer program or, more properly, an interdependent system of programs. While "user friendly," by design, it still produces voluminous amounts of information containing both the detail and the broad overview required to perform effective survivability assessments on aircraft conceptual designs. This version retains the original's modular form, allowing easier "debugging" and possible further modifications. Additionally, menus and subroutines can be changed or new ones added easily without affecting the other aspects of the program.

No project is complete without comment concerning recommendations for further possible improvements. VISAP could encompass other aircraft types, for instance, helicopters and their variety of applications. To enhance industrial usage, current cost of aircraft and the ramifications upon those costs of alternative survivability related components need to be integrated into VISAP.



APPENDIX A  
VARIABLES, SUBROUTINES, AND DEFINITIONS

A.1. ESCORT

A.1.1. Menu 2 Design

A.1.1.1. Menu 21 Aircraft Performance Inputs

TW	thrust to weight ratio
WS	wing loading
WT	ordnance weight
B	wing span
XL	fuselage length
W	fuselage width
EC	engine face to quarter chord
ED	engine diameter
EL	engine length
DL	duct length

A.1.1.2. Menu 22 Susceptibility Features

JAM	jammer number
IRCS	RCS reduction level
IWARN	RWR installed/not installed value
ICHAFF	chaff dispenser installed/not installed value
IRJAM	IR jammer installed/not installed value
IRFLAR	IR flare dispenser installed/not installed value
IRSUP	IR suppression susceptibility value

A.1.1.3. Menu 23 Vulnerability Features

IFS        general fuel system vulnerability value  
IFV        fuel/void interface vulnerability value  
IFE        fuel/engine interface vulnerability value  
IEA        engine arrangement vulnerability value  
IEP        engine protection vulnerability value  
ICS        control system vulnerability value  
ICA        crew arrangement vulnerability value

A.1.2. Menu 3 Combat Scenario

A.1.2.1. Menu 31 Mission Description

XMDA       mission dash altitude  
XMDM       mission dash distance  
XMDD       mission dash Mach number

A.1.2.2. Menu 32 Threat Definition

AAH        air-to-air threat density  
AAD        air-to-air threat diameter  
AAL        air-to-air threat penetration distance  
SAMH       low altitude SAM threat density  
SAMD       low altitude SAM threat diameter  
SAML       low altitude SAM threat penetration distance

A.1.3. Menu 4 Susceptibility Assessment

A.1.3.1. Menu 41 Probability of Detection

PDAAG     P(d) by air-to-air guns  
PDAAM     P(d) by air-to-air IR missiles  
PDSM      P(d) by low altitude SAM

A.1.3.2. Menu 42 Probability of Hit

PHG P(h) by air-to-air guns  
PHM P(h) by air-to-air IR missile  
PHSM P(h) by low altitude SAM

A.1.4. Menu 5 Vulnerability Assessment

A.1.4.1. Menu 51 Vulnerable Area and Probability of Kill Given a Hit

APAAG presented area to air-to-air guns  
AVAAG vulnerable area to air-to-air guns  
PKHAAG P(k/h) by air-to-air guns  
AVAAM vulnerable area to air-to-air IR missile  
PKHAAM P(k/h) by air-to-air IR missile  
VASM vulnerable area to low altitude SAM  
PKHSM P(k/h) by low altitude SAM

A.1.5. Menu 6 Survivability Assessment

A.1.5.1. Menu 61 Probability of Survival

PSAG P(s) vs. air-to-air guns  
PSAM P(s) vs. air-to-air IR missile  
PSSM P(s) vs. low altitude SAM

A.1.5.2. Menu 62 Sortie Analysis

ACR number of aircraft in single sortie  
XNPASS number of targets attacked by aircraft per sortie  
ACR2 number of aircraft ready for next sortie  
TOTSR total sorties flown  
TOTACK total targets attacked  
TOTACL total aircraft lost

TOTACR total aircraft in repair at end  
SORT subroutine to perform sortie analysis

A.1.5.3. Menu 63 Campaign Analysis

ACR1 number of aircraft in campaign  
NSRT number of raids in the campaign  
NS maximum number of sorties for repair

A.1.6. Menu 7 Reassessment

ESRPDS subroutine:  $P(d)$  by low altitude SAM  
ESRPHG subroutine:  $P(h)$  by air-to-air guns  
ESRPHM subroutine:  $P(h)$  by air-to-air IR missile  
ESRPHS subroutine:  $P(h)$  by low altitude SAM  
ESRAVG subroutine:  $A(v)$  and  $P(k/h)$  vs. air-to-air guns  
ESRAVM subroutine:  $A(v)$  and  $P(k/h)$  vs. air-to-air IR missile  
ESRAVS subroutine:  $A(v)$  and  $P(k/h)$  vs. low altitude SAM  
CAMP subroutine to perform campaign assessment

A.1.7. Menu 8 Plotting Routine

N counter for maximum of three plot values  
PKAG  $P(k)$  vs. air-to-air guns array  
PKAM  $P(k)$  vs. air-to-air IR missile array  
PKSM  $P(k)$  vs. low altitude SAM array

A.1.8. Other/Miscellaneous

I1 single digit integer input  
I2 two digit integer input  
V1 real number input  
IJK integer to prevent auto-scroll

KK            general commands comparison array  
K1            Main Menu comparison array  
K2            Menu 2 comparison array  
K3            Menu 3 comparison array  
K4            Menu 4 comparison array  
K5            Menu 5 comparison array  
K6            Menu 6 comparison array  
JJ            Menu 8 comparison array  
K1Q-K9Q      branch command variables  
SRFA          subroutine: alertion factor  
SRFC          subroutine: chaff factor  
ESRWT        subroutine: take off gross weight

A.2. ESCPLT

X0            X-axis points array  
Y0            lower Y-axis values  
Y1            Baseline Design P(k)'s array  
Y2            1st Design P(k)'s array  
Y3            2nd Design P(k)'s array  
IPKRAY       Legend text array

A.3. STRIKE

A.3.1. Menu 2 Design

A.3.1.1. Menu 21 Aircraft Performance Inputs

TW            thrust to weight ratio  
WS            wing loading  
WT            ordnance weight

B wing span  
XL fuselage length  
W fuselage width  
EC engine face to quarter chord  
ED engine diameter  
EL engine length

A.3.1.2. Menu 22 Susceptibility Features

JAM jammer number  
IRCS RCS reduction level  
IWARN RWR installed/not installed value  
ICHAFF chaff dispenser installed/not installed value

A.3.1.3. Menu 23 Vulnerability Features

IFS general fuel system vulnerability value  
IFV fuel/void interface vulnerability value

A.3.2. Menu 3 Combat Scenario

A.3.2.1. Menu 31 Mission Description

XMA mission penetration altitude  
XMD mission penetration distance  
XMM mission penetration Mach number

A.3.2.2. Menu 32 Threat Definition

AAH air-to-air threat density  
AAD air-to-air threat diameter  
SAMH high altitude SAM threat density  
SAMD high altitude SAM threat diameter

A.3.3. Menu 4 Susceptibility Assessment

A.3.3.1. Menu 41 Probability of Detection

PDSM P(d) by high altitude SAM

PDAR P(d) by air-to-air IR missile

A.3.3.2. Menu 42 Probability of Hit

PHSM P(h) by high altitude SAM

PHAR P(h) by air-to-air IR missile

A.3.4. Menu 5 Vulnerability Assessment

A.3.4.1. Menu 51 Vulnerable Area and Probability of Kill Given a Hit

AVAA vulnerable area to air-to-air IR missile

PKHAA P(k/h) by air-to-air IR missile

VASM vulnerable area to high altitude SAM

PKHSM P(k/h) by high altitude SAM

A.3.5. Menu 6 Survivability Assessment

A.3.5.1. Menu 61 Probability of Survival

PSSM P(s) vs. high altitude SAM

PSAR P(s) vs. air-to-air IR missile

A.3.5.2. Menu 62 Sortie Analysis

ACR number of aircraft in single sortie

XINPASS number of targets attacked by aircraft per sortie

ACR2 number of aircraft ready for next sortie

TOTSR total sorties flown

TOTACK total targets attacked

TOTA CL total aircraft lost

TOTACR total aircraft in repair at end

SORT subroutine to perform sortie analysis

#### A.1.5.3. Menu 63 Campaign Analysis

ACR1      number of aircraft in campaign  
XNPASS    number of targets attacked by aircraft in campaign  
NSRT      number of raids in the campaign  
NS        maximum number of sorties for repair

#### A.3.6. Menu 7 Reassessment

SSRPDA    subroutine: P(d) by air-to-air IR missile  
SSRPDS    subroutine: P(d) by high altitude SAM  
SSRPHR    subroutine: P(h) by air-to-air IR missile  
SSRPHS    subroutine: P(h) by high altitude SAM  
SSRAVA    subroutine: A(v) and P(k/h) vs. air-to-air IR missile  
SSRAVS    subroutine: A(v) and P(k/h) vs. high altitude SAM  
CAMP      subroutine to perform campaign assessment

#### A.3.7. Menu 8 Plotting Routine

N         counter for maximum of three plot values  
PKSM      P(k) vs. high altitude SAM array  
PKAR      P(k) vs. air-to-air IR missile

#### A.3.8. Other/Miscellaneous

I1        single digit integer input  
I2        two digit integer input  
V1        real number input  
IJK       integer to prevent auto-scroll  
KK        general commands comparison array  
K1        Main Menu comparison array  
K2        Menu 2 comparison array



K3        Menu 3 comparison array  
K4        Menu 4 comparison array  
K5        Menu 5 comparison array  
K6        Menu 6 comparison array  
JJ        Menu 8 comparison array  
K1Q-K9Q   branch command variables  
SRFA      subroutine: alertion factor  
SRFC      subroutine: chaff factor  
ESRWT     subroutine: take off gross weight

A.4. STRPLT

X0        X-axis points array  
Y0        lower Y-axis values  
Y1        Baseline Design P(k)'s array  
Y2        1st Design P(k)'s array  
Y3        2nd Design P(k)'s array  
IPKRAY    Legend text array  
LABEL     X-axis labels array

A.5. SUPPORT

A.5.1. Menu 2 Design

A.5.1.1. Menu 21 Aircraft Performance Inputs

TW        thrust to weight ratio  
WS        wing loading  
WT        ordnance weight  
B         wing span  
XL        fuselage length

W fuselage width  
ES engine separation  
EC engine face to quarter chord  
ED engine diameter  
EL engine length

A.5.1.2. Menu 22 Susceptibility Features

JAM jammer number  
IRCS RCS reduction level  
IWARN RWR installed/not installed value  
ICHAFF chaff dispenser installed/not installed value

A.5.1.3. Menu 23 Vulnerability Features

IFS general fuel system vulnerability value  
IFV fuel/void interface vulnerability value  
IFE fuel/engine interface vulnerability value  
IEA engine arrangement vulnerability value  
IEP engine protection vulnerability value  
ICS control system vulnerability value  
ICA crew arrangement vulnerability value

A.5.2. Menu 3 Combat Scenario

A.5.2.1. Menu 31 Mission Description

XMA mission loiter altitude  
XMR mission radius of action  
XMT mission time on station

A.5.2.2. Menu 32 Threat Definition

AAAH AAA threat density  
AAAD AAA threat diameter  
SAMH low altitude SAM threat density  
SAMD low altitude SAM threat diameter

A.5.3. Menu 4 Susceptibility Assessment

A.5.3.1. Menu 41 Probability of Detection

PDSM P(d) by low altitude SAM  
PDAR P(d) by AAA radar  
PDAO P(d) by AAA optical

A.5.3.2. Menu 42 Probability of Hit

PHSM P(h) by low altitude SAM  
PHR P(h) by AAA radar  
PHO P(h) by AAA optical

A.5.4. Menu 5 Vulnerability Assessment

A.5.4.1. Menu 51 Vulnerable Area and Probability of Kill Given a Hit

VAAAA vulnerable area to AAA  
PKHAAA P(k/h) by AAA  
VASM vulnerable area to low altitude SAM  
PKHSM P(k/h) by low altitude SAM

A.5.5. Menu 6 Survivability Assessment

A.5.5.1. Menu 61 Probability of Survival

PSSM P(s) vs. low altitude SAM  
PSAR P(s) vs. AAA radar  
PSAO P(s) vs. AAA optical

A.5.5.2. Menu 62 Sortie Analysis

ACR number of aircraft in single sortie  
XINPASS number of targets attacked by aircraft per sortie  
ACR2 number of aircraft ready for next sortie  
TOTSR total sorties flown  
TOTACK total targets attacked  
TOTACL total aircraft lost  
TOTACR total aircraft in repair at end  
SORT subroutine to perform sortie analysis

A.5.5.3. Menu 63 Campaign Analysis

ACR1 number of aircraft in campaign  
XN<sup>2</sup>PASS number of targets attacked by aircraft in campaign  
NSRT number of raids in the campaign  
NS maximum number of sorties for repair

A.5.6. Menu 7 Reassessment

SRPDSM subroutine: P(d) by low altitude SAM  
SRPHSM subroutine: P(h) by low altitude SAM  
SRVASM subroutine: A(v) and P(k/h) vs. low altitude SAM  
SRPHR subroutine: P(h) by AAA radar  
SRPHO subroutine: P(h) by AAA optical  
SRVAAA subroutine: A(v) and P(k/h) vs. AAA  
CAMP subroutine to perform campaign assessment

A.5.7. Menu 8 Plotting Routine

N counter for maximum of three plot values  
PKSM P(k) vs. low altitude SAM array

PKAR P(k) vs. AAA radar  
PKAO P(k) vs. AAA optical

#### A.5.8. Other/Miscellaneous

I1 single digit integer input  
I2 two digit integer input  
V1 real number input  
IJK integer to prevent auto-scroll  
KK general commands comparison array  
K1 Main Menu comparison array  
K2 Menu 2 comparison array  
K3 Menu 3 comparison array  
K4 Menu 4 comparison array  
K5 Menu 5 comparison array  
K6 Menu 6 comparison array  
JJ Menu 8 comparison array  
K1Q-K9Q branch command variables  
SRFA subroutine: alertion factor  
SRFC subroutine: chaff factor  
SSRWT subroutine: take off gross weight

#### A.6. SUPPLT

X0 X-axis points array  
Y0 lower Y-axis values  
Y1 Baseline Design P(k)'s array  
Y2 1st Design P(k)'s array

Y3        2nd Design P(k)'s array  
IPKRAY    Legend text array  
LABEL     X-axis labels array

APPENDIX B

SAMPLE INSTRUCTION MANUAL  
AND ASSESSMENTS

AE 3251  
AIRCRAFT COMBAT SURVIVABILITY

AIRCRAFT SURVIVABILITY DESIGN AND ASSESSMENT  
USING THE  
VOUGHT INTERACTIVE SURVIVABILITY ASSESSMENT PROGRAM  
( VISAP )

NAVAL POSTGRADUATE SCHOOL  
MONTEREY, CALIFORNIA

## INTRODUCTION

The VISAP (Vought Interactive Survivability Assessment Program) was developed at NPS to introduce the student to the survivability decisions and design tradeoffs confronting the designer/analyst of conceptual aircraft. Three specific aircraft types are examined, a Fighter Escort, a Long Range Strike aircraft, and a Close Air Support aircraft. The student is presented with several aircraft performance and design features, potential threats, and vulnerability/susceptibility parameters from which to choose for each aircraft type. Having established a baseline design (either through the default values or by individual design), the student can easily assess the effects of changing one or more design or mission descriptive parameters.

Several measures of the aircraft design's survivability are presented. These include probability of detection ( $P(d)$ ), probability of hit ( $P(h)$ ), and the probability of survival ( $P(d)$ ) against a particular threat for each of the three types of aircraft analyzed. Comparisons of the effectiveness of each design can be obtained through repeated use of the SORTIE and CAMPAIGN analysis models incorporated in the programs. Graphs, of three designs each, may also be obtained for comparison of results.

All required inputs for the analysis are made at a computer terminal. Real time results will appear at the terminal, and hard copy results of each analysis can be sent to the on line printer. Subsequently, plots of loss rate,  $P(k)$ , versus the threat types for each aircraft can be processed at an IBM 3277/Tektronix 618 dual screen terminal. throughout the analysis, default values are used for all calculations unless corrected or updated by the user.



## INSTRUCTIONS

You will need the following items to estimate the survivability and effectiveness of your designs for the three types of aircraft:

1. A computer user number.
2. The ability to LOG ON and operate the IBM 3033 VM system from a terminal.
3. This set of instructions.

The completion of the following instructions causes the VISAP program to execute. VISAP is an interactive program and is self explanatory. Please read the instructions given on the screen carefully. Failure to do so may invalidate your results and terminate the program. Please be sure to enter all variables in the format requested. You are to complete a design evaluation for each type of aircraft. The "HELP" Menus will give you useful information about the program execution and the methodology. It is recommended that you design the Fighter Escort Aircraft (ESCORT) first. It contains the most detail.

In order to access VISAP you must complete the following steps:

1. Turn the terminal on using the red toggle or pull switch on the left hand side.
2. Depress alternately the "RESET" AND "ENTER" keys until the terminal screen is cleared and the message, "CP READ", appears.
3. Enter "L XXXXP", where XXXX refers to your user number (Do not omit the blank space).

4. Enter your password.
5. Enter "CP LINK ++++P 191 195 RR", where ++++ is the user's number on whose disk the programs reside (again do not omit blank spaces).
6. Enter the password "SAP".
7. Enter "ACC 199 B".
8. Enter "VISAP". This calls the exec.
9. Choose and enter one of the aircraft types:  
"ESCORT", "STRIKE", or "SUPPORT".
10. After you have completed your analysis of one aircraft type, you may design another type by exiting the program and then reentering "VISAP" and choosing another type. Requesting printed results of the assessment(s) for each type can be retrieved after you exit that type.

To obtain graphs, you must utilize an IBM 3277/Tektronix 618 dual screen terminal. Follow the VISAP accession procedures, listed above, for steps 1 - 7 as before, then:

8. Enter "DISVIS". This accesses the DISSPLA programs.
9. Choose and enter one of the following: "ESCPLT", "STRPLT", or "SUPPLT" for the Escort, Strike, or Support type aircraft respectively.
10. Push the "HARD COPY" key beneath the large screen for a printout.
11. After receiving a plot for one type of aircraft, you may obtain others by pressing the "ENTER" key and reentering "DISVIS".

## TASKS

You are to complete the following tasks:

1. For each type of aircraft conduct a "BASELINE" (no survivability enhancement features) assessment using the default values.
2. For each type of aircraft, select the survivability features that you want. Then conduct an assessment of that design. What is the weight penalty and how many aircraft are saved in the campaign?
3. For one type of aircraft do a sensitivity study on any three features.

Examples:

- (a) What is the effect of jammer power on the results?
- (b) What is the effect of wing loading on the results?
- (c) What is the effect of the fuel system vulnerability reduction on the results?

Use the plotting procedure to present your results.

4. Comment on whether your studies agree with the theory that you learned in class. Why or why not?
5. Please note any errors or difficulties that you encounter.

## ESCORT INITIAL INPUTS

The following mission, aircraft, and threat parameters are used to conduct the "ESCORT" assessment:

1. Aircraft performance indicators:
  - (a) Thrust to Weight ..... 1.0
  - (b) Wing Loading ..... 70.0 lb/sq ft
  - (c) Ordnance Weight ..... 4000.0 lbs
2. Mission Description:
  - (a) Mission Dash Altitude ..... 10,000.00 ft
  - (b) Mission Dash Mach ..... 0.8
  - (c) Mission Dash Distance ..... 75 miles
3. Threat Definition:
  - (a) Air-to-Air Threat Density ..... 0.01 wpns/sq mi
  - (b) Air-to-Air Threat Diameter ..... 2.0 miles
  - (c) Air-to-Air Penetration Distance ..... 150.0 miles
  - (d) Low Altitude SAM Threat Density ..... 0.0017 wpns/sq mi
  - (e) Low Altitude SAM Threat Diameter ..... 20.0 miles
  - (f) Low Altitude SAM Penetration Distance .. 75.0 miles
4. Sortie and Campaign Analysis:
  - (a) Initial Number of Aircraft ..... 100
  - (b) Number of Raids in Campaign ..... 20
  - (c) Number of Passes per Sortie ..... 1
  - (d) Number of Sorties for Repair ..... 4

## STRIKE INITIAL INPUTS

The following Mission, Aircraft, and Threat parameters are used to conduct the "STRIKE" assessment:

1. Aircraft Performance Indicators:
  - (a) Thrust to Weight ..... 1.0
  - (b) Wing Loading ..... 105.0 lb/sq ft
  - (c) Ordnance Weight ..... 4000.0 lbs
2. Missions Description:
  - (a) Mission Penetration Distance ..... 200.0 miles
  - (b) Mission Penetration Altitude ..... 40000.0 ft
  - (c) Mission Penetration Mach ..... 1.8
3. Threat Definition:
  - (a) Air-to-Air Threat Density ..... 0.01 wpns/sq mi
  - (b) Air-to-Air Threat Diameter ..... 4.0 miles
  - (c) High Altitude SAM Threat Density ..... 0.0017 wpns/sq mi
  - (d) High Altitude SAM Threat Diameter ..... 20.0 miles
4. Sortie and Campaign Analysis:
  - (a) Initial Number of Aircraft ..... 100
  - (b) Number of Raids in Campaign ..... 20
  - (c) Number of Passes per Sortie ..... 1
  - (d) Number of Sorties for Repair ..... 4

## SUPPORT INITIAL INPUTS

The following Mission, Aircraft, and Threat Parameters are used to conduct the "SUPPORT" assessment:

1. Aircraft Performance Indicators:
  - (a) Thrust to Weight ..... 0.55
  - (b) Wing Loading ..... 90.0 lb/sq ft
  - (c) Ordnance Weight ..... 8000.0 lbs
2. Mission Description:
  - (a) Mission Radius of Action ..... 150.0 miles
  - (b) Mission Loiter Altitude ..... 10000.0 ft
  - (c) Mission Time on Station ..... 60.0 min
3. Threat Definition:
  - (a) AAA Threat Density ..... 0.01 wpns/sq mi
  - (b) AAA Threat Diameter ..... 3.0 miles
  - (c) Low Altitude SAM Threat Density ..... 0.0017 wpns/sq mi
  - (d) Low Altitude SAM Threat Diameter ..... 20.0 miles
4. Sortie and Campaign Analysis:
  - (a) Initial Number of Aircraft ..... 100
  - (b) Number of Raids in Campaign ..... 20
  - (c) Number of Passes per Sortie ..... 1
  - (d) Number of Sorties for Repair ..... 4

# FIGHTER ESCORT AIRCRAFT

## Loss Rate VS. Threat Type

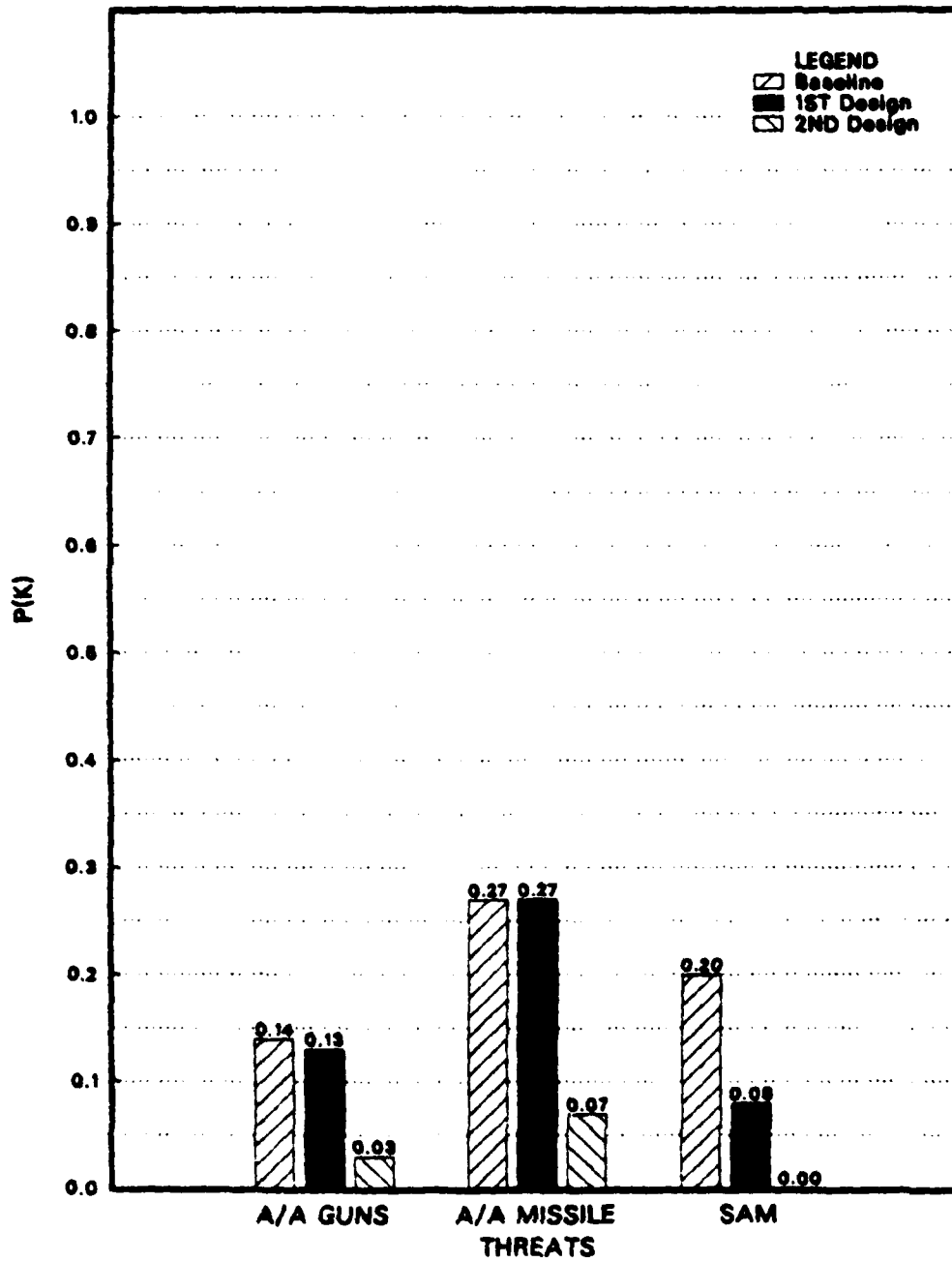


Figure 2. Sample Escort Plot

\*\* FIGHTER ESCORT AIRCRAFT \*\*

```

* PERFORMANCE FEATURES
  THRUST TO WEIGHT      1.00
  ORDNANCE WEIGHT      4000.00
  WING LOADING        70.00

* MISSION PARAMETERS
  DASH ALTITUDE      10000.00
  DASH MACH NBR.     0.80
  DASH DISTANCE      75.00
  A/A DENSITY        0.01
  A/A DIAMETER      2.00
  A/A PENETRATION DIST 150.00
  SAM DENSITY       0.00
  SAM DIAMETER     20.00
  SAM PENETRATION DIST 75.00

* THREAT PARAMETERS

* SUSCEPTIBILITY REDUCTION FEATURES
  JAMMER NUMBER      0
  RCS REDUCTION LEVEL 0
  RADAR WARNING RECEIVER 0
  CHAFF DISPENSER   0
  IR JAMMER         0
  IR FLARE DISPENSER 0
  IR SUPPRESSICN TECHNIQUE 0
  FUEL SYSTEM GENERAL 1
  FUEL/VVOID INTERFACE 1
  FUEL/ENGINE INTERFACE 1
  ENGINE ARRANGEMENT 1
  ENGINE PROTECTION 1
  CONTROL SYSTEM     1
  CREW ARRANGMENT   1

* SURVIVABILITY ASSESSMENT:
  VS A/A GUNS      P(S)  P(D)  P(H)  P(K/H)
  VS A/A MISS     C.86  1.00  0.86  0.17
  VS SAM          C.73  1.00  0.41  0.68
                   C.80  1.00  0.20  0.99

* CAMPAIGN ANALYSIS:
  INITIAL A/C      100.
  PASSES/SORTIE   1.
  A/C READY       43.
  TOTAL TARGETS  1233.
  IN REPAIR      8.
  NUMBER OF RAIDS 20
  SORTIES FOR REPAIR 4
  TOTAL SORTIES  1294.
  TOTAL A/C LOST 49.

BASELINE TOGW    47932.01
ENHANCED TOGW   47932.01
  
```

Figure 3. Sample Escort Baseline Output



\*\* FIGHTER ESCORT AIRCRAFT \*\*

```

* PERFORMANCE FEATURES
  THRUST TO WEIGHT      1.2C
  ORDNANCE WEIGHT      4000.00
  WING LOADING          70.00

* MISSION PARAMETERS
  DASH ALTITUDE      15000.00
  DASH MACH NBR.     0.80
  DASH DISTANCE      75.00
  A/A DENSITY        0.02
  A/A DIAMETER      2.00
  A/A PENETRATION DIST 150.00
  SAM DENSITY       0.00
  SAM DIAMETER     20.00
  SAM PENETRATION DIST 75.00

* THREAT PARAMETERS
  A/A DENSITY        0.02
  A/A DIAMETER      2.00
  A/A PENETRATION DIST 150.00
  SAM DENSITY       0.00
  SAM DIAMETER     20.00
  SAM PENETRATION DIST 75.00

* SUSCEPTIBILITY REDUCTION FEATURES
  JAMMER NUMBER      5
  JCS REDUCTION LEVEL 0
  RADAR WARNING RECEIVER 0
  CHAFF DISPENSER   0
  IR JAMMER DISPENSER 0
  IR SUPPRESSION TECHNIQUE 0

* VULNERABILITY REDUCTION FEATURES
  FUEL SYSTEM GENERAL 8
  FUEL/VOID INTERFACE 1
  FUEL/ENGINE INTERFER 1
  ENGINE ARRANGEMENT 1
  ENGINE PROTECTION 1
  CONTROL SYSTEM     1
  CREW ARRANGMENT    1

* SURVIVABILITY ASSESSMENT:
  VS A/A GUNS        P(S)  P(D)  P(H)  P(K/H)
  VS A/A MISS       0.87  1.00  0.86  0.15
  VS SAM            0.73  1.00  0.40  0.67
                   0.92  0.55  0.09  0.93

* CAMPAIGN ANALYSIS:
  INITIAL A/C      100.
  PASSES/SOFTIE   31.
  A/C READY        976.
  TOTAL TARGETS   12.
  IN REPAIR

  NUMBER CF RAIDS      20
  SORTIES FOR REPAIR  4
  TOTAL SCRTIES      1065.
  TOTAL A/C LCST     57.

  BASELINE TOGW     51861.88
  ENHANCED TOGW     56990.67
  
```

Figure 4. Sample Escort 1st Design Output

\*\* FIGHTER ESCORT AIRCRAFT \*\*

```

* PERFORMANCE FEATURES
  THRUST TO WEIGHT      1.20
  ORDNANCE WEIGHT      10000.00
  WING LOADING        80.00

* MISSION PARAMETERS
  DASH ALTITUDE      15000.00
  DASH MACH NBR.    1.20
  DASH DISTANCE     100.00

* THREAT PARAMETERS
  A/A DENSITY        0.02
  A/A DIAMETER      5.00
  A/A PENETRATION DIST 100.00
  SAM DENSITY       0.00
  SAM DIAMETER     25.00
  SAM PENETRATION DIST 100.00

* SUSCEPTIBILITY REDUCTION FEATURES
  JAMMER NUMBER      5
  RCS REDUCTION LEVEL 6
  RADAR WARNING RECEIVER 1
  CHAFF DISPENSER   1
  IR JAMMER DISPENSER 1
  IR FLARE DISPENSER 1
  IR SUPPRESSICN TECHNIQUE 2

* VULNERABILITY REDUCTION FEATURE
  FUEL SYSTEM GENERAL 8
  FUEL/VCID INTERFACE 8
  FUEL/ENGINE INTERENT 8
  ENGINE ARRANGEMENT 2
  ENGINE PROTECTION   2
  CONTROL SYSTEM      5
  CREW ARRANGMENT     6

* SURVIVABILITY ASSESSMENT:
  VS A/A GUNS        P(S)  P(D)  P(H)  P(K/H)
  VS A/A MISS        C.97  1.00  0.92  0.03
  VS SAM             C.53  1.00  0.17  0.42
                   C.10  0.00  0.29

* CAMPAIGN ANALYSIS:
  INITIAL A/C        100.
  PASSENGER/ SORTIE 49.
  A/C READY TARGETS 1070.
  TOTAL TARGETS     1070.
  IN REPAIR         29.
  NUMBER OF RAIDS   20
  SORTIES FOR REPAIR 4
  TOTAL SORTIES    1202.
  TOTAL A/C LCST   22.

BASELINE TOGW      57463.81
ENHANCED TOGW      84619.50
  
```

Figure 5. Sample Escort 2nd Design Output

# LONG RANGE STRIKE AIRCRAFT

## Loss Rate VS. Threat Type

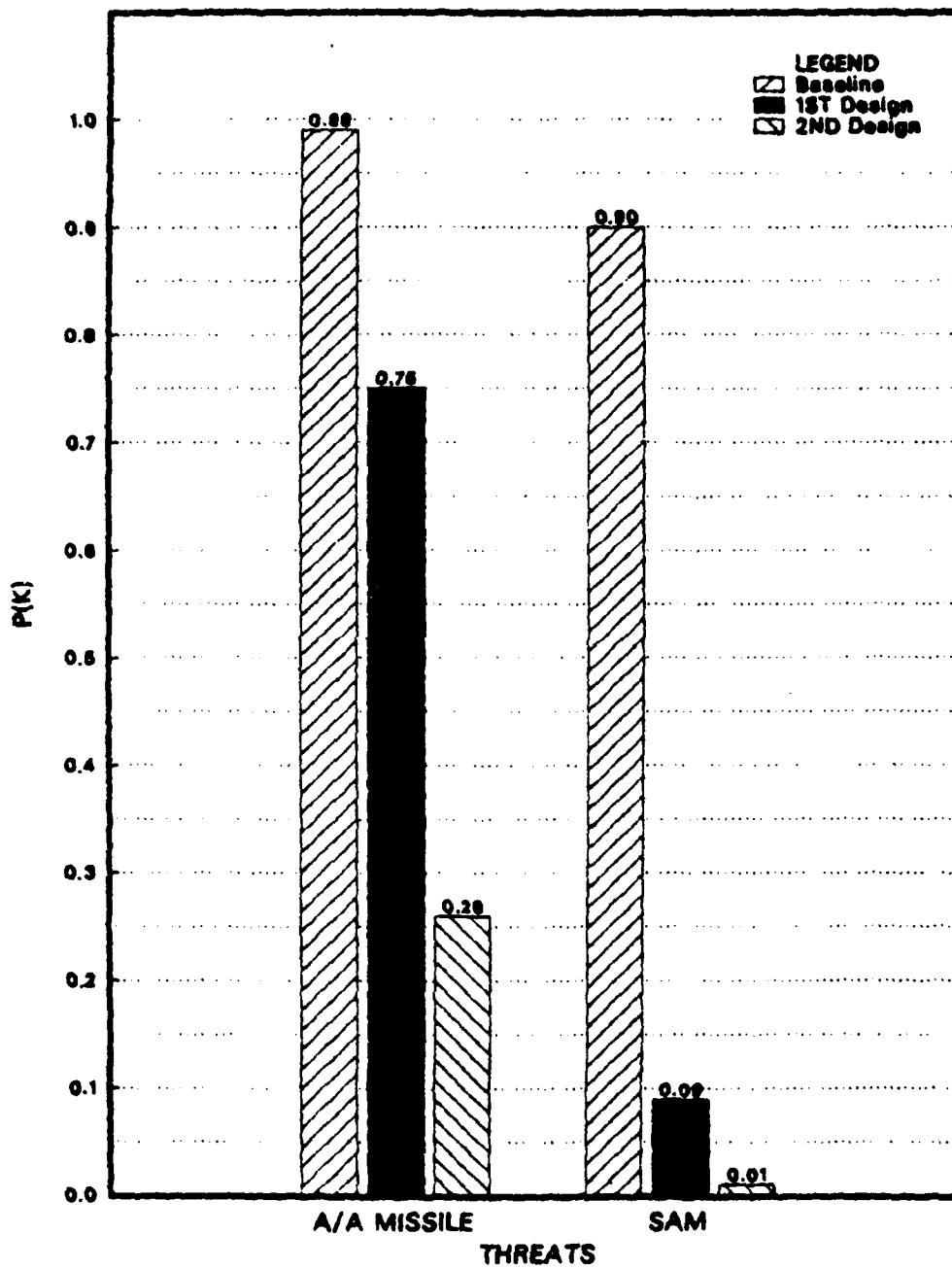


Figure 6. Sample Strike Plot

```

** LCNG RANGE STRIKE AIRCRAFT **

* PERFORMANCE FEATURES
THRUST TO WEIGHT          1.00
ORDNANCE WEIGHT          4000.00
WING LOADING            105.00

* MISSION PARAMETERS
PENETRATION DISTANCE    200.00
PENETRATION ALTITUDE   40000.00
PENETRATION MACH NBR.  1.80
A/A DENSITY             0.01
A/A DIAMETER           4.00
SAM DENSITY            0.00
SAM DIAMETER          20.00

* THREAT PARAMETERS
A/A DENSITY             0.01
A/A DIAMETER           4.00
SAM DENSITY            0.00
SAM DIAMETER          20.00

* SUSCEPTIBILITY REDUCTION FEATURES
JAMMER NUMBER           0
RCS REDUCTION LEVEL    0
RADAR WARNING RECEIVER 0
CHAFF DISPENSER        0
VULNERABILITY REDUCTION FEATURES
FUEL SYSTEM GENERAL    1
FUEL/VOID INTERFACE    1

* SURVIVABILITY ASSESSMENT:
VS A/A MISSILE          P(S)  P(D)  P(H)  P(K/H)
VS HIGH ALT SAM        C.01  C.99  1.00  1.00
                       C.10  C.99  1.00  1.00

* CAMPAIGN ANALYSIS:
INITIAL A/C           100.
PASSES/SORTIE         1.
A/C READY TARGETS    92.
TOTAL TARGETS IN REPAIR 0.
NUMBER OF RAIDS      20
SORTIES FOR REPAIR   4
TOTAL SORTIES        155.
TOTAL A/C LCST      100.

BASELINE TCGW        64071.66
ENHANCED TCGW       64071.66

```

Figure 7. Sample Strike Baseline Output

```

** LCNG RANGE STRIKE AIRCRAFT **

* PERFORMANCE FEATURES
THRUST TO WEIGHT      1.20      WING LOADING      105.00
ORONANCE WEIGHT      4000.00

* MISSION PARAMETERS
PENETRATION DISTANCE  1000.00      A/A DENSITY      0.02
PENETRATION ALTITUDE  4000.00      A/A DIAMETER     4.00
PENETRATION MACH NBR.  1.80        SAM DENSITY      0.00
                                     SAM DIAMETER     20.00

* THREAT PARAMETERS
A/A DENSITY      0.02
A/A DIAMETER     4.00
SAM DENSITY      0.00
SAM DIAMETER     20.00

* SUSCEPTIBILITY REDUCTION FEATURES
JAMMER NUMBER      5
RCS REDUCTION LEVEL  0
RADAR WARNING RECEIVER  0
CHAFF DISPENSER    0

* VULNERABILITY REDUCTION FEATURES
FUEL SYSTEM GENERAL  4
FUEL/VOID INTERFACE  1

* SURVIVABILITY ASSESSMENT:
VS A/A MISSILE      P(S)  P(D)  P(H)  P(K/H)
VS HIGH ALT SAM    0.25  0.76  1.00  0.98
                   0.91  0.10  0.91  0.99

* CAMPAIGN ANALYSIS:
INITIAL A/C        100.
PASSES/SORTIE     1.
A/C READY         0.
TOTAL TARGETS     35.
TOTAL A/C LGST   100.
IN REPAIR        0.
NUMBER CF RAIDS   20
SORTIES FOR REPAIR  4
TOTAL SORTIES    113.
TOTAL A/C LGST   100.

BASELINE TOGW     146325.19      ENHANCED TOGW    159213.56

```

Figure 8. Sample Strike 1st Design Output

\*\* LCNG RANGE STRIKE AIRCRAFT \*\*

```

* PERFORMANCE FEATURES
THRUST TO WEIGHT      1.20      WING LOADING      120.00
ORDNANCE WEIGHT      14000.00

* MISSION PARAMETERS      * THREAT PARAMETERS
PENETRATION DISTANCE    1000.00      A/A DENSITY      0.02
PENETRATION ALTITUDE    60000.00     A/A DIAMETER     5.00
PENETRATION MACH NBR.   2.20         SAM DENSITY      0.00
                                           SAM DIAMETER     25.00

* SUSCEPTIBILITY REDUCTION FEATURES      * VULNERABILITY REDUCTION FEATURES
JAMMER NUMBER          5
RCS REDUCTION LEVEL    8
RADAR WARNING RECEIVER 1
CHAFF DISPENSER        1
                                           FUEL SYSTEM GENERAL  4
                                           FUEL/VOID INTERFACE  6

* SURVIVABILITY ASSESSMENT:
VS A/A MISSILE          P(S)      P(D)      P(H)      P(K/H)
VS HIGH ALT SAM C.99   0.74      0.76      0.40      0.88
                       0.99      0.10      0.19      0.47

* CAMPAIGN ANALYSIS:
INITIAL A/C            100.
PASSES/SORTIE         1.
A/C READY              0.
TOTAL TARGETS         153.
IN REPAIR              0.
NUMBER OF RAIDS       20
SORTIES FOR REPAIR    4
TOTAL SORTIES         221.
TOTAL A/C LCST       100.

BASELINE TCGW      200446.50      ENHANCED TCGW 224027.69

```

Figure 9. Sample Strike 2nd Design Output

# CLOSE AIR SUPPORT AIRCRAFT

## Loss Rate VS. Threat Type

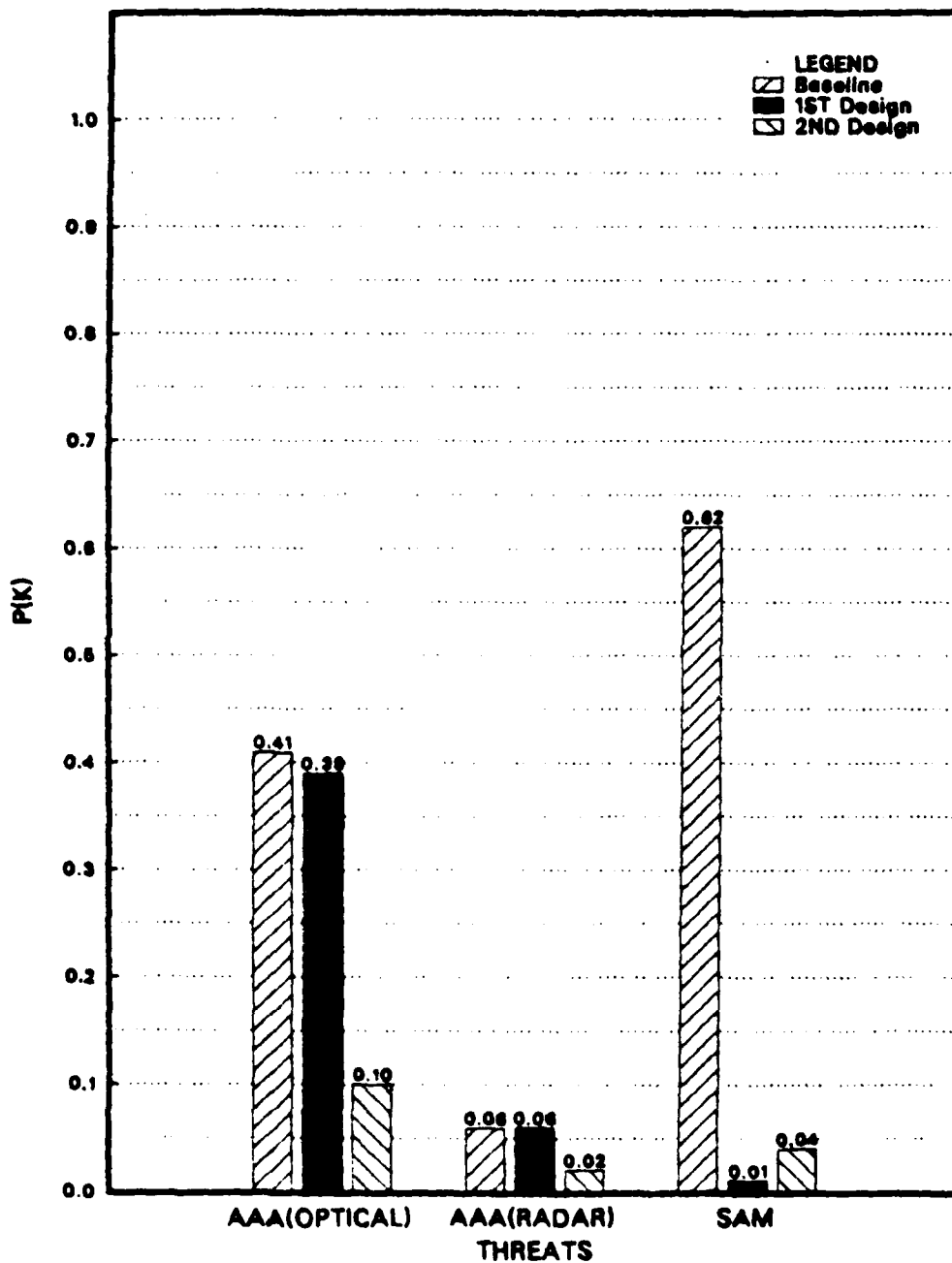


Figure 10. Sample Support Plot

\*\* CLOSE AIR SUPPORT AIRCRAFT \*\*

```

* PERFORMANCE FEATURES
  THRUST TO WEIGHT      0.55      WING LOADING      90.00
  ORDNANCE WEIGHT      8000.00

* MISSION PARAMETERS
  RADIUS OF ACTION      150.00      AAA DENSITY      0.01
  LCITER ALTITUDE      10000.00     AAA DIAMETER     3.00
  TIME ON STATION      60.00       SAM DENSITY      0.00
                                     SAM DIAMETER     20.00

* THREAT PARAMETERS

* SUSCEPTIBILITY REDUCTION FEATURES
  JAMMER NUMBER          0          FUEL SYSTEM GENERAL      1
  RCS REDUCTION LEVEL    0          FUEL/VGID INTERFACE     1
  RADAR WARNING RECEIVER 0          ENGINE ARRANGEMENT      1
  CHAFF DISPENSER        0          ENGINE PROTECTION       1
                                     CONTROL SYSTEM          1
                                     CREW ARRANGMENT        1

* SURVIVABILITY ASSESSMENT:
  VS AAA OPTICAL          P(S)      P(D)      P(H)      P(K/H)
  VS AAA RADAR            0.94      1.00      0.14      0.45
  VS SAM                   0.59      1.00      0.90      0.45
                                     0.38      1.00      0.13      1.00

* CAMPAIGN ANALYSIS:
  INITIAL A/C            100.      NUMBER OF RAIDS          20
  PASSES/SORTIF          1.       SORTIES FOR REPAIR       4
  A/C READY              30.      TOTAL SORTIES           1103.
  TOTAL TARGETS         103%.    TOTAL A/C LCST          65.
  IN REPAIR

  BASELINE TOGW          28945.09     ENHANCED TOGW           28945.09

```

Figure 11. Sample Support Baseline Output



\*\* CLOSE AIR SUPPORT AIRCRAFT \*\*

\* PERFORMANCE FEATURES

THRUST TO WEIGHT	0.65	WING LOADING	90.00
ORDNANCE WEIGHT	8000.00		

\* THREAT PARAMETERS

AAA DENSITY	0.02
AAA DIAMETER	3.00
SAM DENSITY	0.00
SAM DIAMETER	20.00

\* MISSION PARAMETERS

RADIUS OF ACTION	300.00
LCRITER ALTITUDE	10000.00
TIME ON STATION	60.00

\* VULNERABILITY REDUCTION FEATURES

FUEL SYSTEM GENERAL	8
FUEL/VICID INTERFACE	1
FUEL/ENGINE INTERENT	1
ENGINE ARRANGEMENT	1
ENGINE PROTECTION	1
CONTROL SYSTEM	1
CREW ARRANGMENT	1

\* SUSCEPTIBILITY REDUCTION FEATURES

JAMMER NUMBER	5
RCS REDUCTION LEVEL	0
RADAR WARNING RECEIVER	0
CHAFF DISPENSE	0

\* SURVIVABILITY ASSESSMENT:

VS AAA OPTICAL	P(S)	P(C)	P(H)	P(K/H)
VS AAA RADAR	0.94	1.00	0.14	0.44
VS SAM	0.61	1.00	0.89	0.44
	0.99	0.97	0.01	0.98

\* CAMPAIGN ANALYSIS:

INITIAL A/C	100.	NUMBER OF RAIDS	20
PASSES/SCORTIE	97.	SCORTIES FOR REPAIR	4
A/C READY	1974.	TOTAL SCORTIES	1975.
IN REPAIR	0.	TOTAL A/C LCST	1973.

BASELINE TCGW	32200.62	ENHANCED TCGW	34886.78
---------------	----------	---------------	----------

Figure 12. Sample Support 1st Design Output

\*\* CLOSE AIR SUPPORT AIRCRAFT \*\*

\* PERFORMANCE FEATURES  
 THRUST TO WEIGHT 0.65 WING LOADING 100.00  
 GRONANCE WEIGHT 10000.00

\* MISSION PARAMETERS \* THREAT PARAMETERS  
 RADIUS OF ACTION 300.00 AAA DENSITY 0.02  
 ALTITUDE 9000.00 AAA DIAMETER 5.00  
 TIME ON STATION 120.00 SAM DENSITY 0.00  
 SAM DIAMETER 25.00

\* SUSCEPTIBILITY REDUCTION FEATURES \* VULNERABILITY REDUCTION FEATURES  
 JAMMER NUMBER 5 FUEL SYSTEM GENERAL 8  
 RCS REDUCTION LEVEL 3 FUEL/VCID INTERFACE 6  
 RADAR WARNING RECEIVER 1 ENGINE INTERMENT 8  
 CHAFF DISPENSER 1 ENGINE ARRANGEMENT 2  
 CCNTROL SYSTEM 2  
 CREW ARRANGMENT 5 6

\* SURVIVABILITY ASSESSMENT:

VS AAA OPTICAL	P(S)	P(D)	P(H)	P(K/H)
VS AAA RADAR	C.98	1.00	0.14	C.11
VS SAM	C.90	1.00	0.90	0.11
	C.96	0.94	0.21	0.18

\* CAMPAIGN ANALYSIS:

INITIAL A/C	100.	NUMBER OF RAIDS	20
PASSES/SORTIE	1.	SORTIES FOR REPAIR	4
A/C READY	72.	TOTAL SORTIES	1632.
TOTAL TARGETS	1583.	TOTAL A/C LCST	18.
IN REPAIR	11.		

BASELINE TCGW 40858.48 ENHANCED TCGW 46643.16

Figure 13. Sample Support 2nd Design Output

APPENDIX C  
FLOW CHARTS

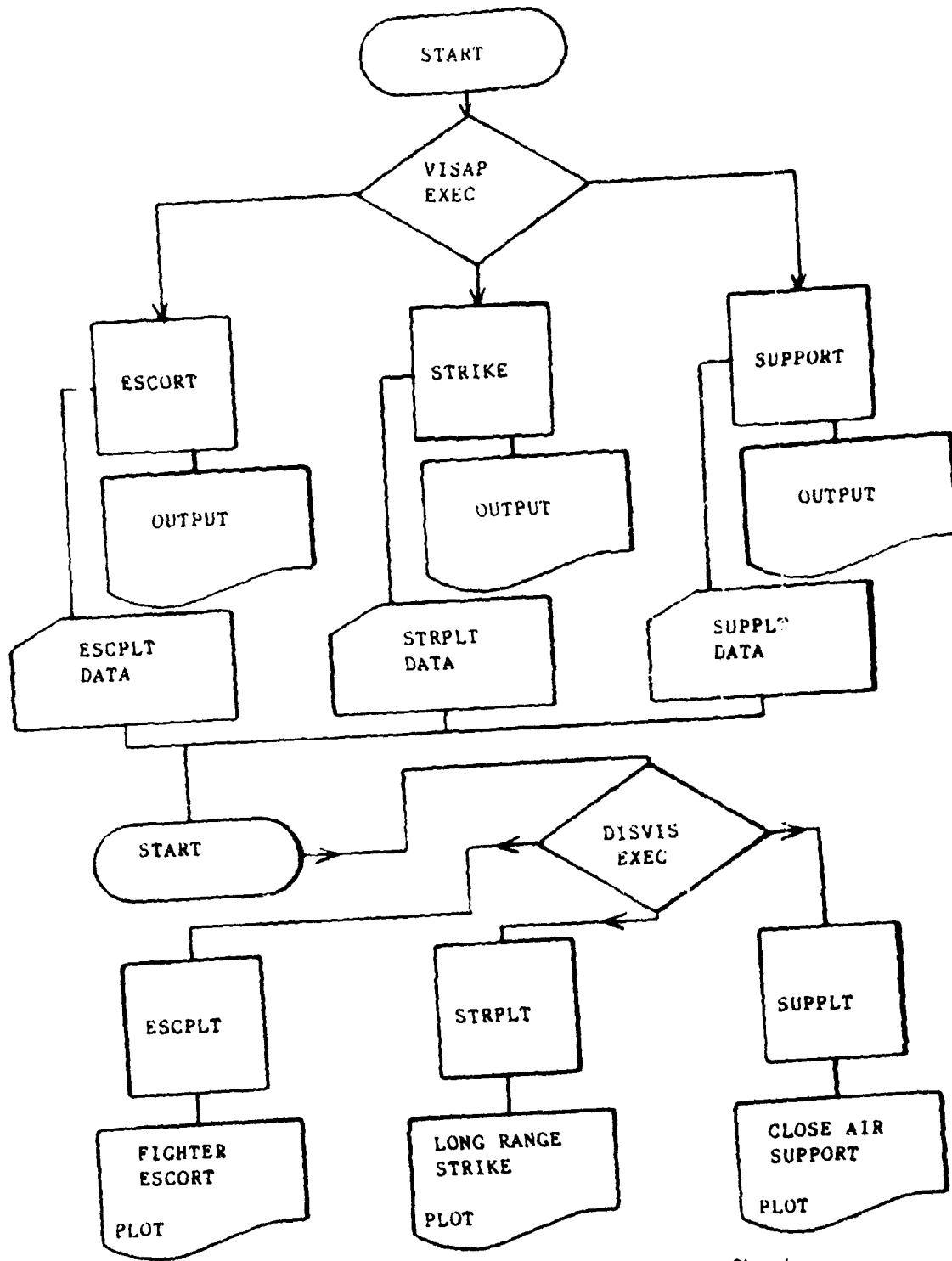


Figure 14. External Program Flow Chart

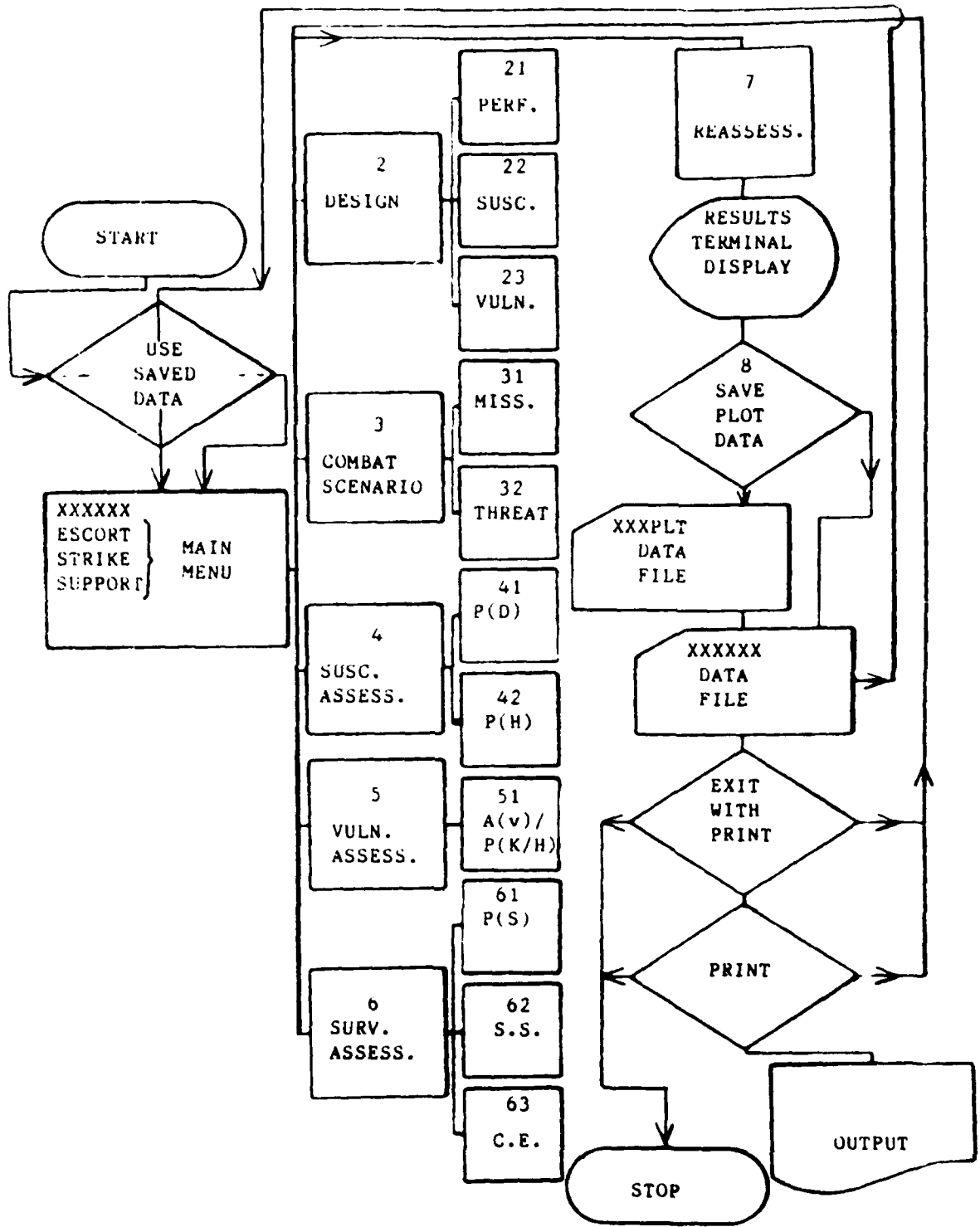


Figure 15. Internal Program Flow Chart

APPENDIX D  
 VISAP AND DISVIS EXECS  
 PROGRAM LISTINGS

```

*****
***
***
***
*****
CMS EXEC TC CONTROL VISAP OPERATION
*****
GLOBAL TXLIB CMSLIB FCRTMCD2 MOD2EEH NONIMSL IMSLSP
FILEDEF 01 DISK ESCORT DATA
FILEDEF 02 DISK STRIKE DATA
FILEDEF 03 DISK SUPPORT DATA
FILEDEF 04 TERMINAL
FILEDEF 06 PRINTER ESCPLT DATA
FILEDEF 07 DISK SUPPLT DATA
FILEDEF 08 DISK SUPPLT DATA
FILEDEF 09 DISK SUPPLT DATA
&TYPE ENTER STRIKE, ESCORT, CR SUPPORT
&READ VARS &1
&LOAD &1 (START
&END

```

```

*****
***
***
***
*****
CMS EXEC TC CONTROL PLOTTING REQUESTS
*****
GLOBAL TXLIB CMSLIB FCRTMCC2 MOD2EEH NONIMSL IMSLSP
FILEDEF 07 DISK ESCPLT DATA
FILEDEF 08 DISK STRPLT DATA
FILEDEF 09 DISK SUPPLT DATA
&TYPE ENTER STRPLT, ESCPLT, CR SUPPLT
&READ VARS &1
&EXEC DISSPLA &1 (START
&END

```





DATA ACRI/100./, NSRI/20/, XNPASS/1./, NS/4/  
 DATA ACRI/2/43.25/, TOTSR/1293.62/, TOTACK/1232.79/, TOTACL/48.71/  
 DATA TCTACR/78.04/, BLTTCGW/47932.61/, TOGW/47932.61/

C- #5 \*\*\*\*\*  
 C \*\*\*\*\*  
 C \*\*\*\*\*  
 C \*\*\*\*\*  
 C \*\*\*\*\*

DATA N/O/ \*\*\*\*\*  
 MAIN MENU DISPLAY \*\*\*\*\*  
 CCNT INUE \*\*\*\*\*  
 CALL FRTCMS('CL:SCRN ') \*\*\*\*\*  
 WRIT(4, 100) ESCRT MENU (1) SELECT A CODE AS FOLLOWS: //  
 +T6, FCR AN, EXPLANATION , T41, HP //  
 +T6, AIRCRAFT DESIGN SELECTION , T41, DE //  
 +T6, COMBAT SCENARIO ASSESSMENT , T41, MS //  
 +T6, SUSCEPTIBILITY ASSESSMENT , T41, SA //  
 +T6, VULNERABILITY ASSESSMENT , T41, VA //  
 +T6, SUPVIVABILITY ASSESSMENT , T41, SV //  
 +T6, TC TRANSFER TO OTHER MENUS , T41, TN //  
 +T6, TC EXIT CR PRINT RESULTS , T41, EX //  
 READ(5, 200) K1C  
 FORMAT(A4)  
 IF(K1C.EQ.K1(1)) GO TO 110  
 IF(K1C.EQ.K1(2)) GO TO 120  
 IF(K1C.EQ.K1(3)) GO TO 130  
 IF(K1C.EQ.K1(4)) GO TO 140  
 IF(K1C.EQ.K1(5)) GO TO 150  
 IF(K1C.EQ.KK(1)) GO TO 9971  
 IF(K1C.EQ.KK(2)) GO TO 998

2000 \*\*\*\*\*  
 C \*\*\*\*\*  
 C \*\*\*\*\*

C- #3 \*\*\*\*\*  
 IF(K1C.EQ.KK(3)) GO TO 1061  
 WRIT(4, 120C)  
 FORMAT(' ', INPUT ERRCR. REPEAT INPUT\*)  
 GO TO 1  
 MENU 2 DESIGN \*\*\*\*\*  
 CALL FRTCMS('CL:SCRN ') \*\*\*\*\*  
 CCNT INUE \*\*\*\*\*

1200 \*\*\*\*\*  
 C \*\*\*\*\*  
 C \*\*\*\*\*  
 C \*\*\*\*\*  
 C \*\*\*\*\*

WRIT(4, 1110)  
 FORMAT(' ', MENU (2) DESIGN, ENTER A CODE AS FOLLOWS: //  
 +T6, FOR AN EXPLANATION INDICATORS , T51, HP //  
 +T6, A/C PERFORMANCE INDICATORS , T51, AP //  
 +T6, SUSCEPTIBILITY FEATURES , T51, SF //  
 +T6, VULNERABILITY FEATURES , T51, VF //  
 +T6, TO RETURN TO MENU (1) , T51, RT //  
 +T6, TO TRANSFER TO OTHER MENUS , T51, TN //





```

C- #3
IF(K4Q.EQ.KK(2)) GC TC 998
WRITE(4,120C)
GO TO 4
*****
C MENU 5 VULNERABILITY ASSESSMENT *****
C ***** VULNERABILITY ASSESSMENT *****
140 CALL FRTCMS('CLRSCRN ')
CCONTINUE
1140 WRITE(4,114C) MENU (5) VULNERABILITY ASSESSMENT'//
FORMAT(4,114C) MENU (5) VULNERABILITY ASSESSMENT'//
+T6, ENTER AN EXPLANATION VS A/A GUN: HP//
+T6, VULN AREA & P(K/H) VS A/A MISSILE: T41, KG//
+T6, VULN AREA & P(K/C) VS A/A MISSILE: T41, KM//
+T6, VULN AREA & P(K/D) VS SAM : T41, KD//
+T6, TC RETURN TO MENU (1) : T41, RT//
+T6, TC TRANSFER TO OTHER MENUS : T41, TN//
READ(5,200,ERR=1061) K54
IF(K5Q.EQ.K5(1)) GC TC 510
IF(K5R.EQ.K5(2)) GC TC 520
IF(K5C.EQ.K5(3)) GC TC 530
IF(K5G.EQ.KK(1)) GC TC 9975
IF(K5Q.EQ.KK(2)) GO TO 998
C- #3
IF(K5C.EQ.KK(4)) GO TO 7
WRITE(4,120C)
GO TO 5
*****
C MENU 6 SURVIVABILITY ASSESSMENT *****
C ***** SURVIVABILITY ASSESSMENT *****
150 CALL FRTCMS('CLRSCRN ')
CCONTINUE
1150 WRITE(4,1150) MENU (6) SURVIVABILITY ASSESSMENT'//
FORMAT(4,1150) MENU (6) SURVIVABILITY ASSESSMENT'//
+T6, ENTER AN EXPLANATION VS A/A GUN: HP, AG//
+T6, P(S) 1:1 A/A (GUNS) : T41, AM//
+T6, P(S) 1:1 A/A (IR MISSILE) : T41, LS//
+T6, P(S) 1:1 LOW ALT SAM : T41, SS//
+T6, SINGLE EVALUATION : T41, CE//
+T6, TC RETURN TO MENU (1) MENUS : T41, RT//
+T6, TC TRANSFER TO OTHER MENUS : T41, TN//
READ(5,200,ERR=1061) K60
IF(K6C.EQ.K6(1)) GO TO 610
IF(K6C.EQ.K6(2)) GO TO 620

```

```

IF(K6C.EQ.K6(3)) GO TO 630
IF(K6Q.EQ.K6(4)) GO TO 640
IF(K6C.EQ.K6(5)) GO TO 650
IF(K6C.EQ.K6(1)) GO TO 9978
IF(K6Q.EQ.K6(2)) GO TO 998

C- #3 IF(K6C.EQ.K6(4)) GO TO 7
WRITE(4,120C)
GO TO 6

C- #3 *****
C ***** MENU 7 REASSESSMENT *****
C ***** CONTINUE *****
CALL ESRPDS (JAM, IRCS, PCSM)
CALL ESRPHG (TW, WS, PHG)
CALL ESRPHM (TW, WS, IRJAM, IRFLAR, IRSUP, PHM)
CALL ESRPHS (IWAR, IFV, ICHAFF, XMDM, WS, PHSM)
CALL ESRVAG (IFS, IFV, IEA, IEP, ICS, ICA, TW, WS, XMDA, XMDM, XMDD, WT,
* AVAAG, APAAG, PKHAAG)
CALL ESRVAM (IFS, IFV, APAAG, AVAAM, PKHAAM)
CALL ESRVAVS (IFS, IFV, APAAG, VASM, PKHSM)

C ***** PSAG = 1. - PDAAG * PHG * PKHAAG *****
C ***** PSAM = 1. - PDAAM * PFM * PKHAAM *****
C ***** PSSM = 1. - PDSM * PHS * PKHSM *****

CALL CAMP (AAL, AAF, AAD, PKHAAG, PSAG, AAL, AAH, AAD, PKHAAM, PSAM,
* SAML, SAMH, SAMD, PKHSM, PSSM, ACRI, NSRT, XNPASS, NS,
* ACR2, TCTSR, TCTACK, TOTACL, TCTACR)

C ***** GO TO 1 *****
C *****

C- #5 *****
C ***** MENU (8) ROUTINE TO GENERATE P(K) VALUES FOR PLOTTING *****
C ***** CONTINUE *****
IF(IN.GET.3) GO TO 999
CALL PRTCMS ('CLASCRN ')
WRITE(4,801)N
FORMAT (1,1,DC YOU WISH TO SAVE P(K) FOR THIS DESIGN? //
* T6, NCTE: YOU HAVE ALREADY CHOSEN , I1, OF THE ,
* T6, NCTE: 3 POSSIBLE DESIGNS /T6, FOR THIS PLOT. //)
801 * * *

```





```

1221 1221 WRITE(4,1221) JAMMERS AVAILABLE //
      FORMAT(0,0)
      +T6,0
      +T6,1
      +T6,2
      +T6,3
      +T6,4
      +T6,5
      +T6,ENTER THE JAMMER NUMBER IN II FORMAT)
      READ(5,1211)II
      JAM=1
      CC CONTINUE
      CALL FRTCMS(,CLKSCRN ,)
      WRITE(4,1222)
      FCFRMT(,0,0)
      1222 RCS REDUCTION LEVELS //
      +T6,0 NO REDUCTION //
      +T6,1 NO REDUCTION //
      +T6,2 PROCEDURE 1 & 2 //
      +T6,3 PROCEDURE 1 & 3 //
      +T6,4 PROCEDURE 1,3, & 5 //
      +T6,5 PROCEDURE 1,3,5, & 6 //
      +T6,6 PROCEDURE 1,4,5,6,7, & 8 //
      +T6,ENTER THE DESIRED RCS LEVEL IN II FORMAT)
      READ(5,1211)II
      IRCS=1
      CC CONTINUE
      1223 RADAR WARNING RECEIVER //
      +T6,0 RADAR WARNING RECEIVER //
      +T6,1 RADAR WARNING RECEIVER //
      +T6,2 RADAR WARNING RECEIVER //
      +T6,3 RADAR WARNING RECEIVER //
      +T6,4 RADAR WARNING RECEIVER //
      +T6,5 RADAR WARNING RECEIVER //
      +T6,6 RADAR WARNING RECEIVER //
      +T6,ENTER THE DESIRED RCS LEVEL IN II FORMAT)
      READ(5,1211)II
      IWRN=1
      CC CONTINUE
      1224 CHAFF DISPENSER //
      +T6,0 CHAFF DISPENSER //
      +T6,1 CHAFF DISPENSER //
      +T6,2 CHAFF DISPENSER //
      +T6,3 CHAFF DISPENSER //
      +T6,4 CHAFF DISPENSER //
      +T6,5 CHAFF DISPENSER //
      +T6,6 CHAFF DISPENSER //
      +T6,ENTER THE DESIRED RCS LEVEL IN II FORMAT)
      READ(5,1211)II
      ICHAF=1
      CC CONTINUE
      1225 IR JAMMER //
      +T6,0 IR JAMMER //
      +T6,1 IR JAMMER //
      +T6,2 IR JAMMER //
      +T6,3 IR JAMMER //
      +T6,4 IR JAMMER //
      +T6,5 IR JAMMER //
      +T6,6 IR JAMMER //
      +T6,ENTER THE DESIRED RCS LEVEL IN II FORMAT)
      READ(5,1211)II
      IIRJ=1
      CC CONTINUE

```

```

226 READ(5,1211)I1
IRJAM=11
GC TO 22C
CCNTINUE
WRITE(4,1226)
FORMAT(,'C' INDICATES NOT INSTALLED "1" INDICATES INSTALLED'//
+T6,'ENTER "0" OR "1" IN I1 FORMAT')
IFFLAR=11
READ(5,1211)I1
GC TO 220
CCNTINUE
CALL FRTCMS('CLPSCRN ')
WRITE(,'')
FORMAT(,'')
+T6,'0 SUPPRESSION TECHNIQUES AVAILABLE'//
+T6,'1 AEROSOL DISPENSER',/
+T6,'2 COLD PLUG',//
+T6,'ENTER THE TECHNIQUE NUMBER IN I1 FORMAT')
READ(5,1211)I1
TPSUP=11
GC TO 22C
C*****
C MENU 23 VULNERABILITY FEATURES*****
C*****
230 CALL FRTCMS('CLPSCRN ')
23 WRITE(4,1230)IFS,IFV,IFE,IEA,IEP,ICS,ICA
FORMAT(,'VULNERABILITY REDUCTION FEATURES'//
+T6,'1 FUEL SYSTEM GENERAL ,I41,I1//
+T6,'2 FUEL/ENGINE INTERFACE ,I41,I1//
+T6,'3 ENGINE ARRANGEMENT ,I41,I1//
+T6,'4 ENGINE PROTECTION ,I41,I1//
+T6,'5 CONTROL SYSTEM ,I41,I1//
+T6,'6 CREW ARRANGEMENT ,I41,I1//
+T6,'7 MINIMUM PROTECTION',/
+T6,'ENTER 0 FOR NO CHANGE ENTER ITS NUMBER IN I1 FORMAT'//
READ(5,1211)I1
IF(I1.EQ.0) GO TC 110
GO TO 232,233,234,235,236,237),I1
1239 WRITE(4,1200)
GO
231 CCNTINUE
CALL FRTCMS('CLPSCRN ')
WRITE(4,1231)

```





```

+T6,12 TWO ENGINES SEPARATED (OVER 2 FT)
+T6,12 ENTER THE DESCRIPTION NUMBER IN I1 FORMAT' )
+T6,12 READ(5,1211)I1
      ICA=I1
      GO TO 230
235  CCNT INDE
      CALL FRTCMS('CLRSRN ')
      WRITE(4,1235)
      WRT('          ENGINE PROTECTION      ' // ' //
+T6,11  NONE
+T6,11  WITH PRCTECTION AND/OR OVER 6 FT OF SEPARATION', //
+T6,12 ENTER THE DESCRIPTION NUMBER IN I1 FORMAT' )
+T6,12 READ(5,1211)I1
      ICA=I1
      GO TO 230
236  CCNT INDE
      CALL FRTCMS('CLRSRN ')
      WRITE(4,1236)
      WRT('          CONTROL SYSTEM POINT FAILURE (SPF) SITES', //
+T6,11  NO BACKUP - UNDER 5 SPF SITES', //
+T6,11  NO BACKUP - UNDER 5 SPF SITES', //
+T6,11  WITH BACKUP - OVER 5 SPF SITES', //
+T6,11  WITH BACKUP - UNDER 5 SPF SITES', //
+T6,11  NO SINGLE POINT FAILURE SITES', //
+T6,11  NO SINGLE POINT FAILURE SITES', //
+T6,11  ENTER THE DESCRIPTION NUMBER IN I1 FORMAT' )
+T6,11 READ(5,1211)I1
      ICA=I1
      GO TO 230
237  CCNT INDE
      CALL FRTCMS('CLRSRN ')
      WRITE(4,1237)
      WRT('          CREW ARRANGEMENT
+T6,11  NO BOTTOM SHIELD FOR PILOT BY ARMOR OR EQUIPMENT', //
+T6,11  NO SIDE SHIELD FOR PILOT BY ARMOR OR EQUIPMENT', //
+T6,11  PARTIAL ARMOR PROTECTION WITH STANDOFF (FRONT AND/OR BOTTC
      &M), //
+T6,11  PARTIAL ARMOR PROTECTION WITH NO STANDOFF (FRONT AND/OR BC
      &TOM), //
+T6,11  FULL ARMOR PROTECTION WITH STANDOFF (FRONT, BOTTOM, AND SI
      &DES), //
+T6,11  FULL ARMOR PROTECTION WITH NO STANDOFF (FRONT, BOTTOM AND
      &SIDES), //
+T6,11  ENTER THE DESCRIPTION NUMBER IN I1 FORMAT' )
+T6,11 READ(5,1211)I1
      ICA=I1
      GO TO 230
C***** MISSION DESCRIPTION *****
C MENU 31

```

```

*****
C* CALL FRICMS('CLRSCRN ')
310 CCNTINUE
311 WRITE(4,1310)XMCA,XMDM,XMDD
    FORMAT(4,1310)
    +T6,'1 MISSION DESCRIPTION',T41,F6.0,' FT. '//
    +T6,'2 MISSION CASH ALTITUDE',T41,F6.4,' MACH. '//
    +T6,'3 MISSION CASH MACH',T41,F6.4,' MACH. '//
    +T6,'4 MISSION CASH DISTANCE',T41,F6.0,' MILES. '//
    +T6,'5 CHANGE THEM ENTER ITS NUMBER IN I1 FORMAT. '//
    +T6,'6 ENTER C FOR NO CHANGE REQUIRED.'
    READ(5,1211)I1
    IF(I1.EQ.0) GO TO 120
    GO TO (311,312,313),I1
1319 WRITE(4,120C)
    GO TO 310
311 CCNTINUE
    WRITE(4,1311)
    FORMAT(4,1311) MISSION CASH ALT RANGE FT.//
    +T6,'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.'
    READ(5,1202)VI
    XMCA=VI
    XMDM=VI
    CCNTINUE
312 GO TO 310
1312 WRITE(4,1312)
    FORMAT(4,1312) MISSION CASH MACH RANGE 0,8 TO 1.2 MACH.//
    +T6,'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.'
    READ(5,1202)VI
    XMDM=VI
    CCNTINUE
313 GO TO 310
1313 WRITE(4,1313)
    FORMAT(4,1313) MISSION CASH DISTANCE NM.//
    +T6,'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.'
    READ(5,1202)VI
    XMDM=VI
    CCNTINUE
C* MENU 32 THREAT DEFINITION *****
C* CALL FRICMS('CLRSCRN ') *****
320 CCNTINUE
321 WRITE(4,1320)AAH,AAC,AAL,SAMH,SAMC,SAML
    FORMAT(4,1320)
    +T6,'1 THREAT DEFINITION',T41,F6.4,' WP/SQ.MI. '//
    +T6,'2 A/A THREAT DENSITY',T41,F6.2,' MI. '//
    +T6,'3 A/A THREAT DIAMETER',T41,F6.2,' MI. '//
    +T6,'4 A/A PENETRATICN DIST',T41,F6.2,' MI. '//
    *****

```

```

+T6,14 SAM THREAT DENSITY ,T41,F6.4, WP/SQ.MI.//
+T6,15 SAM THREAT DIAMETER ,T41,F6.2, MI.//
+T6,16 SAM PENETRATION DIST ,T41,F6.2, MI.//
+T6,17 CHANGE A VALUE ENTER ITS NUMBER IN I1 FORMAT.//
+T6,18 ENTER 0 FOR NO CHANGE REQUIRED.)
READ(5,121)I1
IF(I1.EQ.0) GO TO 120
GO TO (321,322,323,324,325,326),I1
WRITE(4,120)
GO TO 320
CCNTINUE
WRITE(4,1321)
FORMAT(1,1) A/A THREAT DENSITY RANGE 0.0 TO .02 //
ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
READ(5,1202)V1
AAH=V1
GO TO 320
CCNTINUE
WRITE(4,1322)
FORMAT(1,1) A/A THREAT DIAMETER RANGE 0.0 TO 5. //
ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
READ(5,1202)V1
AAC=V1
GO TO 320
CCNTINUE
WRITE(4,1323)XMOD
FORMAT(1,1) A/A PENETRATION DIST RANGE 0.0 TO ,F6.0,
MILES.//
ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
READ(5,1202)V1
AAL=V1
GO TO 320
CCNTINUE
WRITE(4,1324)
FORMAT(1,1) SAM THREAT DENSITY RANGE 0.0 TO .002 //
ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
READ(5,1202)V1
SAMH=V1
GO TO 320
CCNTINUE
WRITE(4,1325)
FORMAT(1,1) SAM THREAT DIAMETER RANGE 0.0 TO 25. //
ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
READ(5,1202)V1
SAMD=V1
GO TO 320
CCNTINUE
WRITE(4,1326)XMOD

```

1329

321

1321

322

1322

323

1323

324

1324

325

1325

326

```

1326 +, MILES,, SAM PENETRATION DIST. RANGE 0.0 TO ,F6.0,
+T6, ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
      READ(5,1202)VI
      SAMPL=VI
      GC TO 320
*****
C MENU 41 PRGB OF DETECTION *****
C *****
410 CALL FRTCMS('CLRSCRN ') *****
41 CCNTINUE *****
      WRITE(4,141) MENU (41) SELECT A CODE AS FOLLOWS:
      FORMAT(4,141) MENU (41) SELECT A CODE AS FOLLOWS:
+T6, P(C) VS A/A (GUNS)
+T6, P(C) VS A/A (IR MISSILE)
+T6, P(D) VS LOW ALTITUDE SAM
+T6, TC RETURN TO MENU (4)
+T6, TC TRANSFER TO OTHER MENUS
      READ(5,200) K70 GO TC 411
      IF(K70.EQ.K6(1)) GO TO 412
      IF(K70.EQ.K6(2)) GO TC 413
      IF(K70.EQ.KK(1)) GC TC 9976
      IF(K70.EQ.KK(2)) GC TC 998
      IF(K70.EQ.KK(4)) GC TO 130
      WRITE(4,141)
      GO TO 41
*****
C PD A/A GUNS *****
C *****
411 CALL FRTCMS('CLRSCRN ') *****
      WRITE(4,141) PD AAG *****
      FORMAT(4,141) THE PROBABILITY OF DETECTION BY A/A (GUNS) IS
+T6, ENTER THE PROBABILITY OF DETECTION BY A/A (GUNS) IS
      READ(5,1201)VI GO TC 410
      IF(VI.EQ.0) GO TO 1413
      WRITE(4,1200)
      GO TO 411
*****
1419 GC TO 411
1413 CCNTINUE
      WRITE(4,1417)
1417 FORMAT(4,1417) PD RANGE 0.0 TO 1.0
+T6, ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
      READ(5,1202)VI
      PD AAG=VI
      GC TO 411

```

```

C*****
C PD A/A IR MISSILE
C*****
412 CALL FRTCMS('CLRSCRN ')
WRITE(4,1415)POAAM
+ F6.4//T6. TC PROBABILITY OF DETECTION BY A/A (MISSILE) IS
+ T6. ENTER Q II FCR NO CHANGE REQUIRED')
READ(5,1211) II GO TC 410
IF(II.EQ.0) GO TC 1416
WRITE(4,1412)
GD CCNTINUE
1418 WRITE(4,1417)
1416 READ(5,1202)V1
PCAAAM=V1
GC TO 412
C*****
C PD VS SAM
C*****
413 CCNTINUE
CALL ESRFDS (JAM IRCS,PDSM)
CALL FRTCMS('CLRSCRN ')
1490 WRITE(4,1491)POSM
1491 FORMAT(' ',T6,' THE COMPUTED PRCBABILITY OF DETECTION BY LOW ALT SA
+ M (RADAR) IS C FCR NO CHANGE REQUIRED')
+ T6. ENTER Q II GO TC 410
READ(5,1211) II GO TC 1492
IF(II.EQ.0) GO TC 1490
WRITE(4,1493)
GD CCNTINUE
1493 WRITE(4,1494)
1494 READ(5,1202)V1
PDSM=V1
GC TO 1450
C*****
C MENU 42 PROB OF HIT
C*****
420 CALL FRTCMS('CLRSCRN ')
42 CCNTINUE
WRITE(4,1420)
1420 FORMAT(' ',MENU (42) SELECT A CODE AS FOLLOWS: '//
+ T6. FOR AN EXPLANATION
+ T6. P(H) VS A/A (GUNS)
+ T6. P(H) VS A/A (IR MISSILE)

```











```

C*****IR MISSILE*****
C(S) A/A IR MISSILE
C*****
620 PSAM = 1 - PDAAM * PHM * PKHAAM
628 CALL FRTCMS(,CLRSCRN, )
WRITE(4,12220)JAM,IFS,IRCS,IFV,IWARN,IFE,ICHAFF,IEA,IRJAM,IEP,
+IRFLAR,ICS,IRSUP,ICA
WRITE(4,1620)PSAM,PDAAM,PHM,PKHAAM
FORMAT(, ,T6, , THE PROB OF SURVIVAL VS A/A (IR MISSILE) '//
+T6, , PS = 1 - PD * PH *
+T9,F6.4,7X,F6.4,3X,F6.4,3X,F6.4//
+T6, ,TC CHANGE THIS VALUE ENTER 1 IN I1 FORMAT. '//
+T6, ,ENTER 0 FOR NO CHANGE REQUIRED. //
READ(5,1659)I1
IF(I1.EQ.0) GO TO 150
GO IN (621,1629),I1
WRITE(4,1200)
CONTINUE
WRITE(4,1611)
READ(5,1657)PDAAM,PHM,PKHAAM
PSAM=1.-PDAAM * PHM * PKHAAM
GO TO 628
C*****
C(S) LCH ALT SAM
C*****
630 PSSM = 1 - PDSM * PHSM * PKHSM
638 CALL FRTCMS(,CLRSCRN, )
WRITE(4,12220)JAM,IFS,IRCS,IFV,IWARN,IFE,ICHAFF,IEA,IRJAM,IEP,
+IRFLAR,ICS,IRSUP,ICA
WRITE(4,1630)PSSM,FDSM,PHSM,PKHSM
FORMAT(, ,T6, , THE PROB OF SURVIVAL VS SAM '//
+T6, , PS = 1 - PD * PH *
+T9,F6.4,7X,F6.4,3X,F6.4,3X,F6.4//
+T6, ,TC CHANGE THIS VALUE ENTER 1 IN I1 FORMAT. '//
+T6, ,ENTER 0 FOR NO CHANGE REQUIRED. //
READ(5,1659)I1
IF(I1.EQ.0) GO TC 150
GO TO (631,1639),I1
WRITE(4,1200)
CONTINUE
WRITE(4,1611)
READ(5,1657)PDSM,PHSM,PKHSM
PSSM=1.-PDSM * PHSM * PKHSM
GO TO 638
C*****SORTIE ANALYSIS*****
C MENU 62

```



```

1650 CALL FRTCMS('CLRSCRN',')
WRITE(4,1650)ACRI,NSRT,XNPASS,NS,PSAG,PSAM,PSSM
FORMAT(1,'MENU (63) CAMPAIGN ANALYSIS',//
+T6,1 AIRCRAFT IN CAMPAIGN .,T46,F6.0/
+T6,2 NUMBER OF RAICES IN CAMPAIGN .,T43,I6/
+T6,3 NUMBER OF PASSES PER SCRTIE .,T44,F6.0/
+T6,4 NUMBER OF SORTIES FOR REPAIR .,T43,I6//
+T6,5 P(S) VS A/(GUNS) ISSILE) .,T48,F6.4/
+T6,6 P(S) VS A/(IR MISSILE) .,T48,F6.4//
+T6,7 P(S) VS LOW ALT SAM .,T48,F6.4//
+T6,8 TO CHANGE A VALUE ENTER ITS NUMBER IN I1 FORMAT.'//
+T6,9 ENTER 0 FOR NO CHANGE REQUIRED.'//
READ(5,1659)I1
FORMAT(11)
IF(I1.EQ.0) GO TO 699
GO TO (651,652,653,654,618,628,638),I1
WRITE(4,1650)ACRI,NSRT,XNPASS,NS,PSAG,PSAM,PSSM
FORMAT(1,'MENU (63) CAMPAIGN ANALYSIS',//
+T6,1 AIRCRAFT IN CAMPAIGN .,T46,F6.0/
+T6,2 NUMBER OF RAICES IN CAMPAIGN .,T43,I6/
+T6,3 NUMBER OF PASSES PER SCRTIE .,T44,F6.0/
+T6,4 NUMBER OF SORTIES FOR REPAIR .,T43,I6//
+T6,5 P(S) VS A/(GUNS) ISSILE) .,T48,F6.4/
+T6,6 P(S) VS A/(IR MISSILE) .,T48,F6.4//
+T6,7 P(S) VS LOW ALT SAM .,T48,F6.4//
+T6,8 TO CHANGE A VALUE ENTER ITS NUMBER IN I1 FORMAT.'//
+T6,9 ENTER 0 FOR NO CHANGE REQUIRED.'//
READ(5,1659)I1
FORMAT(11)
IF(I1.EQ.0) GO TO 699
GO TO (651,652,653,654,618,628,638),I1
WRITE(4,1651)
FORMAT(1,'ENTER NUMBER OF A/C IN REAL NUMBER FORMAT.'//)
READ(5,1657)VI
FORMAT(1,F8.4)
ACRI=VI
GO TO 655
CONTINUE
WRITE(4,1652)
FORMAT(1,'ENTER NUMBER OF PAIDS IN 12 FORMAT.'//)
READ(5,1697)I2
FORMAT(12)
NSRT=I2
GO TO 659
CONTINUE
WRITE(4,1653)
FORMAT(1,'ENTER PASSES PER SCRTIE IN REAL NUMBER FORMAT.'//)
READ(5,1202)VI
XNPASS=VI
GO TO 655
CONTINUE
WRITE(4,1654)
FORMAT(1,'ENTER THE MAX NUMBER OF SCRTIES FOR REPAIR IN I
+2 FORMAT.'//)
READ(5,1697)I2
NS=I2
GO TO 655
CALL FRTCMS('CLRSCRN',')
CALL CAMP( AAL,AAH,AAC,PKHAAG,PSAG;
AAL,AAH,AAC,PKHAAM,PSAM; SAML,SAMH,SAMD,PKHSM,PSSM,
&

```



```

+T6: THE DEFAULT VALUES (BASELINE) ARE ZERO, INDICATING.
+T6: THESE FEATURES ARE INCLUDED. VARY WITH THE THREE.
+T6: VULNERABILITY REDUCTION FEATURES THAT BEST.
+T6: TYPES OF AIRCRAFT. SELECT THOSE FEATURES THAT BEST.
+T6: DESCRIBE YOUR DESIGN. MINIMUM VALUES OF 1 (BASELINE).
+T6: INDICATE ENTER ANY INTEGER TO RETURN TO MENU 2
READ(5,*)IJK
GO TO 110

```

9973 CALL FRTCMS('CLRSCRN')

7973 WRITE(4,7973)

```

+T6: THE COMBAT SCENARIO SECTION IS DIVIDED INTO TWO SUBSECTIONS.
+T6: IN MISSION PROFILE, VALUES ENTERED TO SPECIFICALLY
+T6: DEFINE THE DESIRED MISSION. THE MISSION PARAMETERS ARE DEFINED
+T6: BY THE SELECTION OF AIRCRAFT TYPES. THESE INCLUDE:
+T6: THAT MIGHT BE CONSIDERED AS TACTICS.
+T6: IN THREAT SELECTION, THE THREATS FOR THE FIGHTER ESCORT.
+T6: PARAMETERS ARE ENTERED. THE THREATS FOR THE AIR IR MISSILES, AND
+T6: MISSION ARE: AIR-TO-AIR GUNS, AIR-TO-AIR MISSILES, AND.

```

C- #2

```

+T6: LOW ALTITUDE SAM'S.
+T6: ENTER ANY INTEGER TO RETURN TO MENU 3
READ(5,*)IJK
GO TO 120

```

9974 CALL FRTCMS('CLRSCRN')

7974 WRITE(4,7974)

```

+T6: THE SUSCEPTIBILITY ASSESSMENT SECTION HAS TWO SUBSECTIONS.
+T6: OF THE AIRCRAFT, THE DETECTION IS AFFECTED BY THE
+T6: AND THE SLANT RANGE FROM THE THREAT TO THE A/C AT CPA.
+T6: NOTE THAT ALL AIRCRAFT ARE CONSIDERED TO PASS OVER A POINT.
+T6: THAT IS THE SAME FOR HORIZONTAL DISTANCE FROM THE THREAT.
+T6: AS THE ALTITUDE OF THE AIRCRAFT, THIS MEANS THAT THE
+T6: CPA THE PROBABILITY OF HIT IS DEFINED SEPARATELY FOR EACH.
+T6: AIRCRAFT AND THREAT. HOWEVER, THE FORM IS CONSISTENT, WHERE
+T6: P(H) REFERS TO THE PROBABILITY THAT A NON-MANEUVERING A/C.
+T6: WOULD BE HIT BY THE THREAT. F(M) IS THE MANEUVER FACTOR.
+T6: AND F(C) IS THE COUNTERMEASURE (CHAFF OR FLARE) FACTOR.
+T6: P(H) = P(H) * F(M) * F(C)
+T6: ENTER ANY INTEGER TO RETURN TO MENU 4
READ(5,*)IJK
GO TO 130

```

9975 CALL FRTCMS('CLRSCRN')

7975 WRITE(4,7975)

HELP FOR MENU 5

```

+T6,, THE VULNERABILITY ASSESSMENT SECTION CALCULATES THE //
+T6,, EITHER P(K/H) OR THE AVERAGE VULNERABLE AREA FOR THE //
+T6,, AIRCRAFT VERSUS A THREAT, FOR THE ESCORT AIRCRAFT: //
+T6,, VS A/A GUNS //
+T6,, AV = REGRESSION FORMULA THAT IS A FUNCTION //
+T6,, AP = OF VULNERABILITY FEATURE INPUTS //
+T6,, RE GREGSSION FORMULA THAT IS A FUNCTION //
+T6,, OF MISSION AND A/C PERFORMANCE INPUT DATA. //
+T6,, P(K/H) = AV/AP //
+T6,, VS A/A IR MISSILE //
+T6,, P(K/H) = FROM A TABLE BASED UPON //
+T6,, AP = ASSUMED SAME AS AP VS A/A GUNS (604 SQFT DEFAULT) //
+T6,, AV = P(K/H)*AP //
+T6,, VS LOW ALTITUDE SAM //
+T6,, P(K/H) = FROM A TABLE BASED UPON //
+T6,, AP = ASSUMED SAME AS AP VS A/A GUNS (604 SQFT DEFAULT) //
+T6,, AV = P(K/H)*AP //
+T6,, READ(5,*)IJK //
+T6,, GO TO 140 //
9976 CALL FRTCMS('CLRSCRN ') //
7976 WRITE(4,7976) //
+T6,, THE FOLLOWING METHODS ARE USED FOR THE ESCORT P(D) //
+T6,, VS A/A GUNS/MISSILE //
+T6,, VS P(C) = 1. //
+T6,, VS SAM //
+T6,, VS P(D) = TWO TIMES THE INTEGRAL OF THE GAUSSIAN //
+T6,, PROBABILITY FUNCTION FROM INFINITY TO CPA //
+T6,, ENTER ANY INTEGER TO RETURN TO MENU 41 //
+T6,, READ(5,*)IJK //
+T6,, GO TO 410 //
9977 CALL FRTCMS('CLRSCRN ') //
7977 WRITE(4,7977) //
+T6,, THE FOLLOWING METHODS ARE USED FOR MENU 42 ESCORT P(H) //
+T6,, PH = PROBABILITY THAT A NON-MANEUVERING A/C IS HIT //
+T6,, FA = PROBABILITY THAT THE CREW IS ALERTED AND TAKE EVASIVE AC //
+T6,, VS A/A GUNS //
+T6,, VS P(H) = IR JAMMER FACTOR //
+T6,, VS P(H) = MANEUVER FACTOR //
+T6,, VS P(H) = WING LOADING/100. //
+T6,, VS P(H) = IR SUPPRESSION FACTOR //
+T6,, VS P(H) = IR FLARE FACTOR //
+T6,, VS P(H) = THRUST TO WEIGHT //
+T6,, VS P(H) = 1.24038 * WS - 1.604 * TW //
+T6,, VS P(H) = PM * (1. - (1. - FM)*FA) //
+T6,, VS P(H) = 1. //
79771 WRITE(4,79771) //
+T6,, VS A/A IR MISSILE //

```











```

9202 FORMAT(, , VS A/A GUNS , , 4(F4.2,5X))
9203 FORMAT(, , VS A/A MISS , , 4(F4.2,5X))
9204 FORMAT(, , VS SAM * CAMPAIGN ANALYSIS: , , 17, /
9205 FORMAT(, , INITIAL A/C , , F8.0, , NUMBER OF RAIDS , , 15, /
9206 + + , , PASSES/SORTIE , , F8.0, , SORTIES FOR REPAIR , , F8.0, /
+ + , , A/C READY , , F8.0, , TOTAL SORTIES , , F8.0, /
+ + , , ACTUAL TARGETS , , F8.0, , TOTAL A/C LOST
9207 FORMAT(, , IN REPAIR , , F8.0, , ENHANCED TOGW , , F10.2)
9208 + + , , IN BASELINE TOGW , , F10.2, , 136, , 136, , 136, , 136, /
99999 STOP
END
SUBROUTINE ESRPDS(JAMS, IRCSS, PDF)
*****
***** PROBABILITY OF DETECTION BY LCW ALTITUDE SAM *****
***** DIMENSION H(2,7,6), F(101), PDT(101), XX(101) *****
DATA H/ 3,30,0,9,0,25,8,7,8,24,9,7,6,21,7,2,18,4,5,3,12,2,3,5,
@ 15,0,4,2,13,8,4,0,10,2,3,1,8,5,2,6,7,1,2,1,5,1,1,4,2,4,75,
@ 10,6,3,1,19,7,3,1,7,2,2,1,6,4,1,8,4,9,1,4,3,7,1,1,1,1,5,33,
@ 7,5,2,1,6,9,2,1,5,2,1,4,4,6,1,3,3,4,1,9,6,2,6,1,7,1,0,1,0,0,
@ 4,8,1,1,4,4,3,1,9,6,2,5,1,7,1,2,1,6,3,1,9,5,5,1,3,38,0,1,0,0,
@ 3,6,1,1,1,3,2,1,5,6,2,5,7,1,2,1,1,6,3,1,9,5,5,1,3,38,0,1,0,0,
PDT=0
FOR ALL ALT .LT. 10,000 FT *****
DSR=1.5 ***** CANT USE ZERO AS AN INDEXIES *****
I=IRCSS+1 *****
J=JAMS+1 ***** SELECTS PROPER MEAN AND DEVIATIONS *****
X=H(1,1,J) *****
S=H(2,1,J) *****
IF(S.LT.0) IS=.01 *****
CON1=1.7/(S**2) ***** CONSTANTS FOR EASE OF WRITING *****
CON2=-.5/S**2 *****
XI=X+4 ***** INTEGRATION START AT MEAN + 4 DEVIATIONS *****
STEP=-S/12.5 ***** 100 STEPS IN ITEGRATION *****
F(1)=C.0 ***** INITIAL VALUES TO START INTEGRATION *****
XX(1)=XI *****
PDF=0. ***** INTEGRATION LOOP *****
*****

```

```

00 10 JJ=1,ICO          CLN1*EXP(CON2*(XI-X)**2)
      F(JJ+1)=
      XI=XI+STEP
      AREA=-.5*STEP*(F(JJ+1)+F(JJ))
      PDT(JJ+1)=PD
      IF(XI.LT.CSR)GO TO 5
5      CONTINUE
      XX(JJ+1)=XI
      PD=PD+AREA
10     CONTINUE
      IF(PDF.GT.C.1) GO TO 20
      PDF=.100
      RETURN
      END
C***** SUBROUTINE ESRPHG(TWS,WS,PCAM)
C***** P(H) FOR A/A GUNS
C***** PDAM = 0.
      WSS = WS/100.
      PH = .46168
      FA = 1.
      FM = 1. + 1.24038 * WSS - 1.604 * TW
      XME = 1 + PH * XME
      PCAM = 1 - (1 - FM) * FA
      IF(PCAM.LT..C1)PDAM=.01
      RETURN
      END
C***** SUBROUTINE ESRPHM(TWS,WS,IRJAM,IRFLAR,IRSUP,PDAM)
C***** P(H) FOR A/A IR MISSILE
C***** PDAM = C.
      WSS = WS/100.
      PH = .46168
      IR JAMMER FACTOR
      IF(IRJAM.EQ.1)PH=PH*87
      IF(IRSUP.EQ.1)PH=PH*.011*2**(TW/.2-1.)
      IF(IRFLAR.EQ.1)PH=PH*5
      IF(IRFLAR.EQ.1)PH=PH*5
      FA = 1.
      MANUEVER FACTOR
      FM=-.06056+2.54829*WSS+.06043*TW**2-1.4865?WSS**2-.25379*TW**WSS

```

```

XMF = 1.-(1.-FM)*FA
PCAM = 1.PH*XMF
RETURN
ENC
SUBROUTINE ESRPHS(IWARN,ICHAFF,XMA,WS,PDAM)
*****
C***** PH FCR LOW ALTITUDE SAM *****
C***** PCAM = 0 *****
C***** WSS = WS/100. *****
C***** XXMR = 3.25 *****
C***** PH = .33070 *****
CALL SRFC(ICHAFF,FC) *****
C***** CALL SRFC(MODIFIED FCR, CHAFF) *****
C***** PH = PH * FC *****
CALL SRFA(IWARN,FAI) *****
C***** FM = 1.-1.766*XMA**2+2.9794*WSS**2-XMA**2 *****
C***** IF(FM.LT.C.01)FM = 0.01 *****
C***** XMF = 1.-(1.-FM) * FA *****
PCAM = PH1 * XMF
RETURN
ENC
SUBROUTINE SRFC(ICHAFF,FC)
*****
C***** CHAFF FACTOR *****
C***** REAL PBTSM(17) *****
C***** DATA PBTSM/.00,.19,.35,.49,.6,.68,.74,.8,.83,.86,.9,.92,.935,
A .95,.96,.97,.98/
FC = 0.
PBTSM=0.
IF(ICHAFF.EQ.0) GC TO 10
*****
C***** NBUNDS = 4 *****
PBTSM=PBTSM(NBUNDS+1)
FC = 1. - PBTSM
RETURN
CNC
SUBROUTINE SRFA(XXMRS,IWARN,FAS)
*****
C***** ALERTION FACTOR *****
C***** REAL MRM(28),FVM(28),FESM(12),MRSM(12) *****
C***** DATA MRM/1.,.9995,.97,.91,.82,.69,.53,.35,.22,.137,.065,0./
DATA FVM/0.36,.40,.50,.60,.70,.80,.90,.100,.110,.120,.140./
DATA MRSM/0.1,.1,.194,2.,2.54,3.,3.08,3.48,3.78,4.,4.05,4.23

```



```

AB= 1826.72 -25.1442*B +305.998*A*C +5.49718*A*A*E +.0679002*B*B
& +.000000134505*C*C -.0227707*C*D -.0000872573*C*E
& +.2092535*D*F -.00127764*E*F
AP = ( AF + AB ) / 3.
MES = 16652 + 22272.2*A*D +.000482062*C*C -171.969*B*D
& +.0295446*E*F
& FRS = 67207.8 -163392.*D +7.17324*F -73.9527*A*B +10009.5*A*D
& +144.856*A*E +.0060905*B*C -.051024*B*F +.00000587143*C*C
C- #1 & -----
& - .858783*C*D -.00239668*C*F +105140.*D*D -.0262439*E*F
C*****
FT = SET UP VALUES *****
FS = FLCCAT(IFSS) *****
FV = FLCCAT(IFVS) *****
FE = XFE(IFES) * FT * .001 *****
EAP = XEA(IEAS) * MES * .001 *****
CS = FLCA(IEPS) *****
CA = FLCCAT(ICAS) *****
& AV = CALCULATE VULNERABLE AREA *****
& = 41.56 - 2.244*ALOG(FE) - 4.373*ALOG(FV) - 4.732*ALOG(FS)
& + 5.09*ALCG(CS) + 5.946*ALOG(EA) - 2.491*ALCG(CA)
& + 16.44*ALCG(FT*.001) - 47.503 * ALOG(EP)
C*****
PKH CALCULATE P(K/H) *****
RETURN AV/AV *****
END *****
SUBROUTINE ESRVMS(IFSS,IFVS,AP,AVMS,PKHSMS) *****
*****VULNERABLE AREA AND P(K/H) VS AA IR MISSILE *****
*****SET UP VALUES *****
F1 = 0. *****
F2 = 0. *****
F3 = 0. *****
IF(IFSS.GE.31).AND.(IFSS.NE.3).AND.(IFSS.NE.6) F1 = 1. *****
IF((IFVS.EQ.3).OR.(IFVS.EQ.4).OR.(IFVS.EQ.6)) F2 = 1. *****
IF((IFVS.EQ.5).OR.(IFVS.EQ.6)) F3 = 1. *****
IF(CALCULATE PK/H VS SAM ***** F4 = 1. *****
ITF = INT(F1+F2+F3+F4) *****
GO TC (IC,20,30,40),ITF *****
GC PKHSMS = .675 *****
GO TC 50 *****
GC PKFSMS = F1*.671 + F2*.673 + F3*.648 + F4*.560 *****
GO TC 50 *****
GC PKFSMS = F1*F2*.670 + F1*F3*.640 + F1*F4*.610 +

```



```

8      F2*F3*.644 + F2*F4*.616 + F3*F4*.485
30     GO TC 50
        PKHSMS = F1*F2*F3*.637 + F1*F2*F4*.606 +
        F1*F3*F4*.448 + F2*F3*F4*.460
40     GO TC 50
        PKFSMS = .423
50     CONTINUE
C      CALCULATE VULN AREA VS SAM *****
C      AVMS = PKHSMS*AP *****

```

```

RETURN
END
SURROUTINE ESRV(S,IFSS,IFVS,AP,AV,PKHSAM) *****
C      ***** VULNERABLE AREA VS LOW ALTITUDE SAM *****
C      ***** SET UP VALUES *****
C      F1 = 0. *****
C      F2 = 0. *****
C      F3 = 0. *****
C      F4 = 0. *****

```

```

C      IF(IFSS.GE.3) .AND. (IFSS.NE.3) .AND. (IFVS.NE.6) ) F2 = 1. *****
C      IF((IFVS.EQ.3) .OR. (IFVS.EQ.4) .CR. (IFVS.EQ.6) ) F3 = 1. *****
C      IF((IFVS.EQ.5) .PK/H VS SAM *****
C      IF(CALCULATE(F1+F2*F3+F4) *****
C      ITF = INT((F1+F2*F3+F4)/ *****
C      GC TC (10,20,30,40),ITF *****
C      GO PKFSAM = .9934 *****

```

```

1      GO TC 50
10     PKFSAM = F1*.940 + F2*.964 + F3*.898 + F4*.789
20     GO TC 50
        PKHSAM = F1*F2*.928 + F1*F3*.711 + F1*F4*.880 +
        F2*F3*.783 + F2*F4*.904 + F3*F4*.747
30     GO TC 50
        PKHSAM = F1*F2*F3*.633 + F1*F2*F4*.861 +
        F1*F3*F4*.518 + F2*F3*F4*.494
40     GO TC 50
        PKFSAM = .292
50     CONTINUE
C      CALCULATE VULN AREA VS SAM *****
C      AV = PKHSAM*AP *****

```

```

RETURN
END
SUBROUTINE SORT( *****
C      ***** SORTIE ANALYSIS *****
C      ***** XL1,XH1,D1,FKH1,FS1, XL2,XH2,C2,PKH2,PS2, XL3,XH3,D3,PKH3,PS3, *****

```

ACR1, NSRT, XNPAS, NS, ACR2, TCTSF, TOTACK, TOTACL, TGTACR)

ACR1, NSRT, XNPAS, NS,

ACR2 = 0

TCTSF = 0

TOTACK = 0

TGTACR = 0

TCTACR = 0

ACCLAM = 0

W1 = XL1 \* XH1 \* D1 / 100

W2 = XL2 \* XH2 \* D2 / 100

W3 = XL3 \* XH3 \* D3 / 100

PSM1 = PS1 \*\* W1

PSM2 = PS2 \*\* W2

PSM3 = PS3 \*\* W3

PH1 = (1. - PSM1) / PKH1

PH2 = (1. - PSM2) / PKH2

PH3 = (1. - PSM3) / PKH3

\*\*\*\*\*THREAT 1\*\*\*\*\*

H1 = ACR1 \* PH1

A1 = H1 - XK1

H2 = (ACR1 - H1) \* PH2

XK2 = H2 \* PKH2

A2 = H2 - XK2

H3 = (ACR1 - H1 - H2) \* PH3

XK3 = H3 \* PKH3

A3 = H3 - XK3

ACOVER = TARGET \*\*\*\*\*

ATAC = ACOVER \* XNPAS \*\*\*\*\*

\*\*\*\*\*THREAT 1\*\*\*\*\*

H4 = ACCOVER \* PH1

XK4 = H4 \* PKH1

A4 = H4 - XK4

H5 = (ACCOVER - H4) \* PH2

XK5 = H5 \* PKH2

A5 = H5 - XK5

\*\*\*\*\*THREAT 3\*\*\*\*\*

H6 = (ACCOVER - H4 - H5) \* PH3

XK6 = H6 \* PKH3

A6 = H6 - XK6

-----

#1

AD-A128 203

CORRECTIONS AND IMPROVEMENTS TO THE INTERACTIVE  
COMPUTER PROGRAM FOR THE (U) NAVAL POSTGRADUATE SCHOOL  
MONTEREY CA R M HILL MAR 83

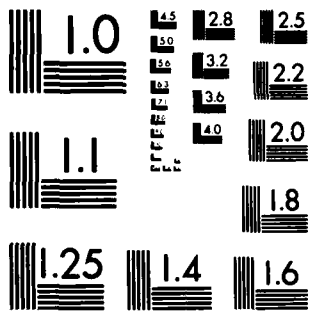
212

UNCLASSIFIED

F/G 9/2

NL

END  
DATE  
FILMED  
0-01  
DTIC



MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A



```

H3 = (ACR(I)-H1-H2) * PH3
XK3 = H3 * PKH3
A3 = H3 - XK3
*****OVER TARGET*****
ACOVER = ACR(I)-H1-H2-H3
ATAC = ACOVER * XNPAS
*****EGRESS*****
H4 = ACOVER * PH1
XK4 = H4 * PKH1
A4 = H4 - XK4
H5 = (ACOVER-H4) * PH2
XK5 = H5 * PKH2
A5 = H5 - XK5

```

```

C- #1 -----
H6 = (ACOVER-H4-H5) * PH3
XK6 = H6 * PKH3
A6 = H6 - XK6

```

```

*****TCTALS FOR SORTIE*****
ACNHT = ACR(I)-H1-H2-H3-H4-H5-H6
ACDAM = A1 + A2 + A3 + A4 + A5 + A6
ACKIL = XK1 + XK2 + XK3 + XK4 + XK5 + XK6
*****FOR NEXT SORTIE*****
TOTACR = TOTACR+ACDAM
ACROUT = TOTACR/FLOAT(NS)
TOTACR = TOTACR-ACROUT
ACR(I+1) = ACNHT + ACROUT
TOTSR = TOTSR + ACR(I)
TOTACK = TOTACK + ATAC
TOTACL = TOTACL + ACKIL

```

```

10 CONTINUE
RETURN

```

```

SUBROUTINE ESRWT(SPANI,ML,EC,ED,EL,TW,WS,WT,XL,ICA,ICS,IEA)
*****
TCGW CALCULATION SUBROUTINE *****
IEP,IFE,IFS,IFV,JAM,IRCS,XMDA,XMDM,XMDD,IRJAM,IRSUP,IWARN,
I,CHAF,IF,IFLAR,BL,TCGW,TOGW)
REAL A,B,C,D,E,F,G,H,I,J
A = WS
B = XMDA
C = XMDM
D = XMT
E = WT
F = 0.
G = 0.
H = 0.

```

```

C*****
I = 0.
J = 0.
BLTOGW = .223614E+05 - .324752E+01*H - .949382E+05*J - .25362*A*C
@+.241287E+05*A*D + .214144E+03*A*B + .36411E+01*A*H + .51498E+05*A*I
@+.130783E+01*C*D - .4681582E-02*C*E - .237024E+05*B*C*G + .6914731E+05*D*I
@+.148915E+06*D*F + .612094E-01*E*G + .366191E+01*D*H - .507909E-02*E*I
@+.378503E+03*E*H + .359190E+03*E*J - .917125E-03*F*I + .205186E+07*F*G
@+.165260E+08*G*H + .111890E+03*G*I - .265987E+07*G*J - .592433E+07*G*J
@+.637399E-04*H*I + .620790E+01*H*J
C*****

```

```

C*****
A/C TOGW CF DESIGN WITH SURVIVABILITY ENHANCEMENT
C*****
THE FOLLOWING ASSUMPTION MADE: 23 MM
ALL SELF-SEALING TANKS HAVE EQUAL VOLUME
DUAL SUMP TANKS I HAVE EQUAL VOLUME
EACH TANK HOLDS 1/7 OF TOTAL VOLUME
INTERNAL FOAM USE VICE EXTERNAL INERTING
FIRE EXTINGUISHING VICE EXTERNAL INERTING
C*****
IF (JAM.EQ.0) GO TO 40
G = .8675 * WS / BLTOGW
CONTINUE
C*****
TEMP FIX CN G *****
C*****
IF (IRSUP.NE.2) GC TO 50
G = .05
J = .05
CONTINUE
C*****

```

```

C*****
WEIGHT INCREASE CALCULATIONS
C*****
FR = 672078E+05*A*B + .163392E+06*A*D + .717334E+01*F - .782443E+05*I + 01*A*H
@+.4739527E+02*A*B + .160095E+05*A*D + .144856E+03*A*H - .519256E+05*I + 01*A*H
@+.238850E+05*A*I + .431972E+05*C*H + .609094E-02*B*C - .510240E-01*B*F
@+.265054E+02*C*G + .587143E-05*C*I + .858783E+00*C*D - .239668E-02*C*D
@+.265054E+02*D*G + .698039E+05*D*H + .951930E+00*C*I + .105140E+06*D*I
@+.265667E+03*E*H + .170282E+08*G*H + .262439E-01*E*F + .646932E+04*E*G
@+.281936E+07*G*H + .184747E-04*F*I + .314433E+01*H*J - .232114E+07*G*I
C*****
XNT = 2.
C*****

```

```

IF((IFS.EQ.1).OR.(IFS.EQ.3).OR.(IFS.EQ.6)) XNT = 0.
IF((IFS.EQ.2) XNT = 1.
WSSP = 1.49*(2.2*8./7.-1.)*(1./7.)*.75*(FR/6.6)**.64*XNT**.11
C***** WEIGTH INCREASE DUE TO INTERNAL FOAM *****
WF = 0
IF((IFV.EQ.4).OR.(IFV.EQ.6)) WF = .0186 * FR/6.6
C***** WEIGTH INCREASE DUE TO FIRE EXTINGUISHING *****
WFE = 0.4/3. * (ED + 1.) * ED * EL
IF((IFV.EQ.5).OR.(IFV.EQ.6)) WFE = 10.5 * XV**.26
C***** WEIGTH INCREASE DUE TO DUCT PROTECTION *****
XND = 1.
WBB = 0.
IF(IFE.EQ.2) XNC = 2.
IF(IFE.EQ.5) XND = .5
IF(IFE.EQ.2).OR.(IFE.EQ.4).OR.(IFE.EQ.6)) WBB = 7.6 * XS
C***** AD *****
AD = 0
IF((ICA.EQ.2) AD = 10.
IF((ICA.EQ.3).OR.(ICA.EQ.4)) AD = 30.
IF((ICA.EQ.5) AD
WARMT INCREASE DUE TO ENGINE SEPERATION*****
WEIGTH INCREASE DUE TO ENGINE SEPERATION*****
XEB = 0
IF(IEA.EQ.1) XEB = 0.
IF(IEA.EQ.2) XEB = 4.
IF(IEP.EQ.2) XEB = 6.
XA = FD/2. * (BLTOGW - FR) * .5
XH = 1.23 * XA
XN = 11.
WENG = 2000.
C- #1 WES = 1.264 + .034 * XA * XT * XH)*(WENG * XN * XA * XEB *.1.0E-10)
C***** WEIGTH INCREASE DUE TO RAM *****
XS = 0
IF((IRCS.EQ.1) XS = 10.
IF((IRCS.EQ.2).OR.(IRCS.EQ.3)) XS = 20.
IF((IRCS.EQ.4)) XS = 60. + BLTOGW/WS *.69
IF((IRCS.EQ.5)) XS = 75. + BLTOGW/WS *.69
IF((IRCS.EQ.6) XS = 23.8
WGRAM = XT * XS * 23.8
WEIGTH INCREASE DUE TO REDUNDANT CONTROLS *****
BACKUP = 0.
IF((ICS.EQ.3).CR.(ICS.EQ.4)) BACKUP = 1.
IF((ICS.EQ.5) + EC + SPAN + XL / 2.
XLGP = BACKUP * (2.207 * XLGP - 4.79)
WRED

```



```

C***** WEIGHT INCREASE DUE TO RWR *****
WEI = 0.
IF(IWAPN.EQ.1) WEW = 50.
C***** WEIGHT INCREASE DUE TO RADAR JAMMER *****
WJW = 0.
IF(JAM.EC.1) WJW = 80.
IF(JAM.EC.2) WJW = 100.
IF(JAM.EC.3) WJW = 200.
IF(JAM.EC.4) WJW = 500.
IF(JAM.EC.5) WJW = 1000.
C***** WEIGHT INCREASE DUE TO CHAFF DISPENSER *****
WEC = 0.
IF(CHAFF.EQ.1) WCC = 86.
C***** WEIGHT INCREASE DUE TO IR FLARE DISPENSER *****
WFD = 0.
IF(IRFLAR.EQ.1) WFC = 86.
C***** WEIGHT INCREASE DUE TO SUBMERGED STORE *****
WFSOR = 0.
IF(IRCS.EQ.5) OR.(IRCS.EQ.6) WSOR = 1.13 * WT/100.
C***** WEIGHT INCREASE DUE TO COOLED IR PLUG *****
WPLG = 0.
IF(IRSUP.EQ.2) WPLG = .01012 * EC**2
C***** WEIGHT INCREASE DUE TO AEROSOL INJECTOR *****
WAI = 0.
IF(IRSUP.EQ.1) WAI = 200.
C***** WEIGHT INCREASE DUE TO IR JAMMER *****
WIRJ = 0.
IF(IRJAM.EQ.1) WIRJ = 200.
C***** TOTAL WEIGHT INCREASE *****
H = WSSP+WFE+WBE+WARM+WES+WRAM+WRED+WEN+WJW+WCD+WFD
+WSCR+WPLG+WAI+WIRJ
C***** TOTAL TOGW OF ENHANCED A/C *****
TOGA = .223614E+05 - .324752E+05 * H - .949382E+05 * J - .25362 * A * C
+.241278E+05 * A * C + .214144E+03 * B * C - .46815E+03 * B * C - .237024E+05 * B * G + .691265E-05 * C * C
+.1185228E+01 * C * D - .4482582E-02 * C * D * F + .6122094E+07 * D * G - .2966972E+02 * C * H - .553071E+05 * D * I
+.148915E+06 * D * J - .2339190E+03 * E * F + .366191E+01 * D * H - .366191E+01 * E * G + .907909E-02 * E * H
+.178503E+03 * E * G + .3359190E+03 * E * J - .917125E-03 * E * F - .205186E+04 * E * G
+.1637399E-04 * H * H + .111890E+03 * G * H - .265987E+07 * G * I + .592438E+07 * G * J
-.637399E-04 * H * H
C***** RETURN
C***** END

```



```

CALL BLBAR (, LABEL, Y0, Y1, 3)
CALL VBARS (, LABEL, Y0, Y2, 3)
CALL VBARS (, LABEL, Y0, Y3, 3)
CALL HEIGHT (.05)
CALL DCT
CALL GRID (0, 2)
CALL RESET (, COT, )
CALL HEIGHT (, 10)
CALL BLOFF (, )
CALL MAXLINE (, IPKRAY, 400, 40)
CALL LINES (, P(ASELINE), , IPKRAY, 1)
CALL LINES (, 1ST C(ESIGN), , IPKRAY, 2)
CALL LINES (, 2ND C(ESIGN), , IPKRAY, 3)
CALL LEGEND (, IPKRAY, 3, 4.5, 7.6)
CALL ENDPL (, )
CALL CCNEPL
CALL STOP
END

```



```

C- #5 -----
DIMENSION KK(4),K1(6),K2(3),K3(2),K4(2),K5(2),K6(4),
      *      JJ(2),PKAR(3),PKSM(3)
C
DATA K1/DE,MS,SA,VA,SV,MM/
DATA K2/AP,SF,VF/
DATA K3/MP,TH/
DATA K4/FD,PH//
DATA K5/KH,KD//
DATA K6/AR,HS,SS,CE//
DATA KK/HP,TN,EX,RT//
C- #5 DATA JJ/Y, N/

```

```

C- #3 -----
C ***** TO SAVE DATA *****
C *****

```

```

1010 CALL FRTCMS('CLRSCRN ')
      WRITE(4,1010)
      FORMAT(1,'DATA MODE SELECTION, ENTER A CODE AS FOLLOWS: /',
      *T6,'IF THIS IS YOUR FIRST THROUGH STRIKE OR IF YOU WISH /',
      *T6,'TO USE THE DEFAULT VALUES/PARAMETERS ENTER: 0 /',
      *T20,'WARNING //',
      *T6,'DO NOT ENTER 1 IF THIS IS YOUR FIRST RUN OR IF YOU HAVE /',
      *T6,'ERASED YOUR STRIKE DATA FROM YOUR DISK-- //')

```

```

C
1011 READ(4,1011)I1
      FORMAT(1)
      IF(I1.EQ.0)GC TO 1021
      IF(I1.EQ.1)GC TO 1022
      CONTINUE
1022 READ(2,1012)TW,WS,WT,EC,EL,ES,JAM,IRCS,IWARN,ICHAFF,IFS,
      *      IFV,XMD,XMA,XMP,DSR,SAMH,AAH,AAAD,
      *      PDAR,PCSM,PHAR,PHSM,AVAA,PKHAA,VASM,PKFSM,PSAR,
      *      PSSM,ACR,XINPAS,ACR1,ANSRT,XNPASS,NS,
      *      ACR2,TOTSR,TOTACK,TOTACL,TOTACR,BLTOGW,TOGW
      *      READ(1,5612.4)
      *      GO TO 1
      *      CONTINUE

```

```

C *****
1021 DATA TW/1.0,WS/105./,WT/4000./,EC/3.5/,EL/9.0/,ES/7.0/
      *      DATA JAM/0/,IRCS/0/,IWARN/0/,ICHAFF/0/
      *      DATA IFS/1/,IFV/1/
      *      DATA XMD/200./,XMA/4000./,XMM/1.8/,DSR/9.31/
      *      DATA SAMH/.0017/,SAMD/20./,AAH/.01/,AAD/4./
      *      DATA PDAR/.5900/,PDSM/.9947/
      *      *****

```



```

+T6, A/C PERFORMANCE INDICATORS
+T6, SUCEPTIBILITY FEATURES
+T6, VULNERABILITY FEATURES
+T6, TC RETURN TO MENU (1)
+T6, TC TRANSFER TO OTHER MENUS
READ(5,2000) K20
IF(K20.EQ.K2(1)) GC TC 210
IF(K20.EQ.K2(2)) GC TC 220
IF(K20.EQ.K2(3)) GC TC 230
IF(K20.EQ.KK(1)) GC TC 9972
IF(K20.EQ.KK(2)) GC TC 998

C- #3 IF(K2C.EC.KK(4)) GO TC 7
WRITE(4,120C)
GO
C*****
C COMBAT SCENARIC
C*****
120 CALL FRTCMS('CLRSCRN ')
CONTINUE
WRITE(4,1120)
FORMAT('1, EXPLANATION
+T6, MISSION PROFILE
+T6, THREAT RETURN TO MENU (1)
+T6, TC TRANSFER TO OTHER MENUS
READ(5,2000) K30
IF(K30.EQ.K3(1)) GO TC 310
IF(K30.EQ.K3(2)) GO TC 320
IF(K30.EQ.KK(1)) GO TC 9973
IF(K30.EQ.KK(2)) GO TC 998

C- #3 IF(K3C.EQ.KK(4)) GO TC 7
WRITE(4,1200)
GO
C*****
C SUCEPTIBILITY ASSESSMENT
C*****
130 CALL FRTCMS('CLRSCRN ')
CONTINUE
WRITE(4,1130)
FORMAT('1, MENU (4) SUCEPTIBILITY ASSESSMENT,
+T6, ENTER A CODE AS FOLLOWS:
+T6, FOR AN EXPLANATION
+T6, PROBABILITY OF DETECTION
+T6, PROBABILITY OF HIT
+T6, TO RETURN TO MENU (1)

```

```

+T6, TC TRANSFER TO OTHER MENUS , T51, TN //
READ(5,2000) K4C
IF(K4C.EQ.K4(1)) GO TO 410
IF(K4C.EQ.K4(2)) GC TC 420
IF(K4C.EQ.K4(1)) GO TO 9974
IF(K4C.EQ.K4(2)) GO TO 998
-----
IF(K4C.EQ.K4(4)) GO TO 7
WRITE(4,1200)
GO TO 4

```

C- #3

```

*****
C***** VULNERABILITY ASSESSMENT *****
C***** VULNERABILITY ASSESSMENT *****
140 ***** CALL FRTCMS('CLRSCRN ') *****
5 ***** CCNT INUE *****
1140 WRITE(4,1140) MENU (5) VULNERABILITY ASSESSMENT, //
FORMAT(1,1) A CODE AS FOLLOWS: //
+T6, ENTER AN EXPLANATION VS A/A MISSILE, HP, T51, KH //
+T6, VULN AREA & P(K/H) VS A/A MISSILE, T51, KH //
+T6, VULN AREA & P(K/D) VS SAM, T51, KH //
+T6, TO RETURN TO MENU (1), T51, KH //
+T6, TO TRANSFER TO OTHER MENUS, T51, TN //
READ(5,2000,ERR=1061) K5C
IF(K5C.EQ.K5(1)) GO TC 510
IF(K5C.EQ.K5(2)) GC TC 520
IF(K5C.EQ.K5(1)) GC TC 9975
IF(K5C.EQ.K5(2)) GC TO 998
-----
IF(K5C.EQ.K5(4)) GO TO 7
WRITE(4,1200)
GO TO 5

```

C- #3

```

*****
C***** SURVIVABILITY ASSESSMENT *****
C***** SURVIVABILITY ASSESSMENT *****
150 ***** CALL FRTCMS('CLRSCRN ') *****
6 ***** CCNT INUE *****
1150 WRITE(4,1150) MENU (6) SURVIVABILITY ASSESSMENT, //
FORMAT(1,1) A CODE AS FOLLOWS: //
+T6, ENTER AN EXPLANATION VS A/A MISSILE, HP, T51, KH //
+T6, P(S) 1:1 A/A (RADAR MISSILE), T51, KH //
+T6, P(S) 1:1 HIGH ALT SAM, T51, KH //
+T6, SINGLE SCRTEE EVALUATION, T51, KH //
+T6, CAMPAGN RETURN TC MENU (1), T51, KH //
+T6, TO TRANSFER TO OTHER MENUS, T51, TN //
READ(5,2000,ERR=1061) K6Q

```







```

1212 GC TO 210
1213 CCNTINUE
1214 WRITE(4,10,' WS RANGE ALLCWD IS 90. TO 120.'//
1215 FCRMAT(10,' VALUE IN REAL NUMBER FORMAT.')
```

```

1216 +T6, 'ENTER(5,1202)V1
1217 WS=V1
1218 GO TO INUE 210
1219 CCNTINUE
1220 WRITE(4,1204)
1221 FCRMAT(10,' WT RANGE ALLCWD IS 1000. TO 14000.'//
1222 +T6, 'ENTER(5,1202)V1
1223 WT=V1
1224 GC TO 210
1225 CCNTINUE
1226 *****
1227 C#***** SUSCEPTIBILITY FEATURES *****
1228 C#***** CALL FRTCMS('CLRSCRN ') *****
1229 CCNTINUE
1230 WRITE(4,1220)JAM,IRCS,IMARN,ICHAFF A CODE AS FOLLOWS:'//
1231 FCRMAT(10,' JAMMER NUMBER LEVEL :T51,I1//
1232 +T6, '1 RADAR REDUCTION FEATURES :T51,I1//
1233 +T6, '2 RCS REDUCTION LEVEL :T51,I1//
1234 +T6, '3 CHAFF DISPENSER :T51,I1//
1235 +T6, '4 CHAFF DIES NOT INSTALLED "1" INDICATES INSTALLED'//
1236 +T6, '5 TO CHANGE A VALUE ENTER ITS NUMBER IN I1 FORMAT'//
1237 +T6, '6 TO ENTER 0 FOR NO CHANGE REQUIRED.')
```

```

1238 READ(5,1211)I1
1239 IF(I1.EQ.0) GO TC 110
1240 GO TO (221,222,223,224),I1
1241 WRITE(4,1200)
1242 GO TO CCNTINUE
1243 CCNTINUE
1244 CALL FRTCMS('CLRSCRN ')
1245 WRITE(4,1221) JAMMERS AVAILABLE '//'
1246 +T6, '00 :T51, I1//
1247 +T6, '50 WATTS :T51, I1//
1248 +T6, '100 WATTS :T51, I1//
1249 +T6, '200 WATTS :T51, I1//
1250 +T6, '500 WATTS :T51, I1//
1251 +T6, '1000 WATTS :T51, I1//
1252 +T6, 'ENTER THE JAMMER NUMBER IN I1 FCRMAT.')
```

```

1253 +T6, '5,1211)I1
1254 JAME=I1

```





```

311          CCNTINUE 1311)
1311          WRITE(4,0) PENETRATION DISTANCE RANGE 100. TO 1000. NM.//
          +T6, ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
          READ(5,1202)VI
          XMC=VI 310
          GO TO 310
312          CCNTINUE 1312)
1312          WRITE(4,0) PENETRATION ALTITUDE RANGE 40000 TO 60000. FT.//
          +T6, ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
          READ(5,1202)VI
          XMA=VI
          DSR = XMA/4256.4
          GO TO 310
313          CCNTINUE 1313)
1313          WRITE(4,0) PENETRATION MACH RANGE 1.4 TO 2.2 MACH.//
          +T6, ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
          READ(5,1202)VI
          XMH=VI
          GO TO 310
C*****
C          MENU 32 THREAT DEFINITION *****
C*****
320          CALL FRTCMS('CLRSCRN ') *****
32          CCNTINUE *****
1320          WRITE(4,1) AAD,SAMH,SAMC *****
          FORMAT(1) MENU (32) *****
          +T6, 1 THREAT DEF *****
          +T6, 2 A/A THREAT DENSITY *****
          +T6, 3 A/A THREAT DIAMETER *****
          +T6, 4 SAM THREAT DENSITY *****
          +T6, 5 SAM THREAT DIAMETER *****
          +T6, 6 CHANGE THEM ENTER ITS NUMBER.// *****
          READ(5,1211)I1 *****
          IF(I1.EQ.0) GO TC 120 *****
          GO TO (321,322,323,324),I1 *****
1329          WRITE(4,1) *****
          GO *****
321          CCNTINUE 1321) *****
1321          WRITE(4,0) A/A THREAT DENSITY RANGE 0.0 TO .02 *****
          +T6, ENTER THE NEW VALUE IN REAL NUMBER FORMAT.) *****
          READ(5,1202)VI *****
          XAH=VI *****
          GO TO 32C *****

```

```

322 CCNTINUE
1322 WRITE(4,1322)
      +T6, 'A/A THREAT DIAMETER RANGE 0.0 TO 5.0' //
      +T6, 'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.' //
      +T6, 'READ(5,1202)V1'
      AAC=V1
      GO TO 320
323 CCNTINUE
1323 WRITE(4,1323)
      +T6, 'SAM THREAT DENSITY RANGE 0.0 TO .002' //
      +T6, 'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.' //
      +T6, 'READ(5,1202)V1'
      SAMH=V1
      GO TO 320
324 CCNTINUE
1324 WRITE(4,1324)
      +T6, 'SAM THREAT DIAMETER RANGE 0.0 TO 25.0' //
      +T6, 'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.' //
      +T6, 'READ(5,1202)V1'
      SAMD=V1
      GO TO 320
C*****
C***** MENU 41 PRCB OF DETECTION *****
C***** CALL FRTCMS('CLRSCRN') *****
410 CCNTINUE
411 WRITE(4,1410)
      +T6, 'MENU (41) SELECT A CODE AS FOLLOWS:' //
      +T6, 'FOR AN EXPLANATION MISSILE) :T51,HR //
      +T6, 'P(D) VS A/A (RADAR MISSILE) :T51,AR //
      +T6, 'P(C) VS HIGH ALT. SAM :T51,HS //
      +T6, 'TC RETURN TC MENU (4) :T51,RT //
      +T6, 'TC TRANSFER TO OTHER MENUS :T51,TN //
      READ(5,2000) K70
      IF(K70.EQ.K6(1)) GO TC 411
      IF(K70.EQ.K6(2)) GO TC 413
      IF(K70.EQ.KK(1)) GO TC 9976
      IF(K70.EQ.KK(2)) GO TC 998
      WRITE(4,1200)
      GO TO 41
C*****
C***** PD A/A RADAR MISSILE *****
C***** CALL FRTCMS('CLRSCRN') *****
411 CALL FRTCMS('CLRSCRN')
4119 WRITE(4,1411)
4141 PDAR
1411 FORMAT('I',T6,'THE COMPUTED PROBABILITY OF DETECTION BY A/A (RADAR

```

```

+J MISSILE IS 'F6.4//T6,' TO CHANGE THIS VALUE ENTER 1'//
+T6, ' ENTER 0 FOR NO CHANGE REQUIRED')
READ(5,1201)I1
IF(I1.EQ.0) GO TC 410
IF(I1.EQ.1) GO TC 1413
WRITE(4,120C)
GO
1419
1413 CCNTINUE(1417)
WRITE(4,120C)
1417 +T6, ' ENTER THE NEW VALUE IN REAL NUMBER FORMAT.'//
      PD RANGE C.0 TO 1.C.0'//
      PCAR=V1
      READ(5,1202)I1
      GO TO 4119
C*****
C PD VS SAM
C*****
413 CCNTINUE
CALL SSRPDS(JAM,IRCS,DSR,PDSM)
CALL ALL FRTCMS('CLRSRN ')
1490 WRITE(4,1451)PDSM
1491 FORMAT('I,X,F6.4//T6,' THE COMPUTED PRCB OF DETECTION BY THE HIGH ALT SAM
+T6, ' ENTER 1'// TO CHANGE THIS VALUE ENTER 1'//
      READ(5,1211)I1
      IF(I1.EQ.0) GO TC 410
      IF(I1.EQ.1) GO TC 1492
      WRITE(4,120C)
      GO
1493
1492 CCNTINUE
WRITE(4,1417)
READ(5,1202)I1
PCSM=V1
GO TO 1490
C*****
C MENU 42 PRCB OF HIT
C*****
420 CALL FRTCMS('CLRSRN ')
CCNTINUE
42
WRITE(4,1420)
FORMAT('I,X,F6.4//T6,' MENU (42) SELECT A CODE AS FOLLOWS:'//
+T6, 'FCR AN EXPLANATION MISSILE) 'T51,HP'//
+T6, 'P(H) VS A7A (RADAR MISSILE) 'T51,PAR'//
+T6, 'P(H) VS HIGH ALT. SAM 'T51,HS'//
+T6, 'TO RETURN TO MENU (4) 'T51,RT'//
+T6, 'TO TRANSFER TO OTHER MENUS 'T51,TN'//
      READ(5,20C0)K89
      IF(K8C.EQ.X&(1)) GC TC 421

```





```

*****
C MENU 51 VULN. AREA / P(K/H) VS A/A RACAR MISSILE *****
C *****
510 *****
CONTINUE SPAAVA(IFS,IFV,AVAA,PKHAA)
CALL FRTCMS(1,1512)AVAA,PKHAA
WRITE(4,1512)THE COMPUTED VULN AREA VS A/A MISSILE IS
FORMAT(1,1512)THE P(K/H) IS ,T51,F6.4,//
+T51,FC CHANGE FCN NC THESE VALUES ENTER 1,//
+T6,FC CHANGE FCN NC CHANGE REQUIRED,
READ(5,1202)I1
IF(I1.EQ.0) GO TO 14C
WRITE(4,120C)
GO TO 1511
1514 *****
CONTINUE
WRITE(4,1514)
FORMAT(1,1514) VULN. AREA RANGE 00.0 TO 600.0 //
+T6,FC CHANGE FCN NC VALUE IN REAL NUMBER FORMAT,
READ(5,1202)I1
AVAA=VI
PKHAA=V1
GO TO 1511
1515 *****
VULN. AREA / P(K/H) VS HIGH ALT SAM *****
C *****
520 *****
CONTINUE
CALL FRTCMS(1,1522)VASM,PKHSM
WRITE(4,1522)THE COMPUTED VULN AREA VS SAM IS ,T51,F6.0/T6,
FORMAT(1,1522)THE P(K/H) IS ,T51,F6.4,//T6,FC CHANGE THESE VALUES ENTER 1,
+T6,FC CHANGE FCN NC CHANGE REQUIRED,
READ(5,1202)I1
IF(I1.EQ.0) GO TO 140
WRITE(4,120C)
GO TO 1521
1524 *****
CONTINUE
WRITE(4,1524)
FORMAT(1,1524) VULN. AREA RANGE 00.0 TO 600.0 //
+T6,FC CHANGE FCN NC VALUE IN REAL NUMBER FORMAT,
READ(5,1202)I1
VASM=VI
PKHSM=V1
GO TO 1521
C *****

```

```

C MENU 61 P(S) A/A RACAR MISSILE *****
C***** PCAR * PHAR * PKHAA *****
610 CALL FRTCMS(CLRSCRN) *****
618 WRITE(4,2220)JAM,IFS,IRCS,IFV,IMARN,ICHAFF *****
      FORMAT(0,0,0,0,SUSCEPTIBILITY REDUCTION FEATURES ', *****
+T40,0,0,VULNERABILITY REDUCTION FEATURES ' // *****
+T42,0,JAMMER NUMBER GENERAL *****
+T42,0,FUEL REDUCTION LEVEL *****
+T42,0,FUEL/VOIC INTERFACE *****
+T42,0,RADAR WARNING RECEIVER *****
+T42,0,CHAFF DISPENSER *****
+T42,0,WRITE(4,1610)PSAR,PCAR,PHAR,PKHAA *****
      FORMAT(0,0,0,0,THE PROB OF SURVIVAL VS AA(RADAR)) // *****
1610 +T6,F6,4,7X,F6,4,3X,F6,4,3X,F6,4 // *****
+T6,0,TC CHANGE THIS VALUE ENTER 1 IN I1 FORMAT // *****
+T6,0,ENTER 0 FOR NO CHANGE REQUIRED *****
      READ(5,1659)I1 GO TC 150 *****
      IF(I1.EQ.0) GO TC 150 *****
      GO TO (611,1619),I1 *****
1619 WRITE(4,1620) *****
      GC TO 618 *****
      CCNTINUE 1611) *****
611 WRITE(4,1619)I1 *****
1611 +T6,0,ENTER 0,16,0,ENTER P(D),P(H),P(K/H) IN REAL NUMBER FORMAT // *****
      READ(5,1657)PDAR,PHAR,PKHAA *****
      PSAR=1,PCAR*PHAR*PKHAA *****
      GC TO 618 *****
C***** P(S) HIGH ALT SAM *****
C***** P(S) ***** PHSM * PKHSM *****
630 PSSM = 1,PCSM * PHSM * PKHSM *****
638 CALL FRTCMS(CLRSCRN) *****
      WRITE(4,2220)JAM,IFS,IRCS,IFV,IMARN,ICHAFF *****
      FORMAT(0,0,0,0,SUSCEPTIBILITY REDUCTION FEATURES ', *****
      PSAR=1,PCAR*PHAR*PKHAA *****
      PHSM=1,PCSM*PHSM*PKHSM *****
1630 +T6,F6,4,7X,F6,4,3X,F6,4,3X,F6,4 // *****
+T6,0,TC CHANGE THIS VALUE ENTER 1 IN I1 FORMAT // *****
+T6,0,ENTER 0 FOR NO CHANGE REQUIRED *****
      READ(5,1659)I1 GO TC 150 *****
      IF(I1.EQ.0) GO TC 150 *****
      GO TO (631,1639),I1 *****
1639 WRITE(4,1620) *****
      GC TO 638 *****

```



```

GO TO 15C
C*****
C MENU 63 *****
C*****
650 *****
659 *****
1650 *****
      MS(,CLRSCRN ,)
      CALL FRT(50)ACR1,NSRT ,XNPASS,NS,PSAR,PSSM
      WRITE(4,16 ,)MENU (63) ,CAMPAIGN ANALYSIS ,//
      FORMAT( ,1 ,)A/CRAFT IN CAMPAIGN ,T43,I6/
      +T6, ,1 ,)NUMBER OF RAIDS IN CAMPAIGN ,T46,F6.0/
      +T6, ,2 ,)NUMBER OF PASSES PER SCRTIE ,T44,I6/
      +T6, ,3 ,)NUMBER OF SORTIES FOR REPAIR ,T43,I6//
      +T6, ,4 ,)P(S) VS A/MISSILE ,T48,F6.4/
      +T6, ,5 ,)P(S) VS A/HEIGHT SAM ,T48,F6.4/
      +T6, ,6 ,)TO CHANGE A VALUE ENTER ITS NUMBER IN I1 FORMAT. //
      +READ(5,1659)II
1659 *****
      IF(11,FQ,51,652,653,654,618,638),I1
      GO TO 699
1658 *****
      WRITE(4,1658)
      GO TO 65C
651 *****
      CCNTINUE(4,1651)
      WFORMAT( ,0 ,T6, ,)ENTER NUMBER OF A/C IN REAL NUMBER FORMAT. //
1651 *****
      READ(5,1657)VI
      FORMAT( ,F8.4)
1657 *****
      ACRI=VI
      GO TO 659
652 *****
      CCNTINUE(4,1652)
      WFORMAT( ,0 ,T6, ,)ENTER NUMBER OF RAIDS IN I2 FORMAT. //
1652 *****
      FCRMAT( ,0 ,T6, ,)ENTER NUMBER OF RAIDS IN I2 FORMAT. //
1697 *****
      FCRMAT( ,12)
      NSRT=12
      GO TO 655
653 *****
      CCNTINUE(4,1653)
      WFORMAT( ,0 ,T6, ,)ENTER PASSES PER SCRTIE IN REAL NUMBER FORMAT. //
1653 *****
      FOR READ(5,1202)VI
      XNPASS=VI
      GO TO 659
656 *****
      CCNTINUE(4,1656)
      WFORMAT( ,0 ,T6, ,)ENTER MAX NUMBER OF SORTIES FOR REPAIR IN I2 FO
1656 *****
      +RMT READ(5,1697)I2

```



```

+T6.. IMPACT CF THESE VALUES MAY BE FOUND IN THE P(H) SECTION.//
+T6.. OF SUSCEPTIBILITY EVALUATION. FEATURES OF THE DESIGN ARE.//
+T6.. SUSCEPTIBILITY SECTION. THESE INCLUDE JAMMER SIZE.//
+T6.. ENTERED IN THIS SECTION. CHAFF DISPENSE AND RADAR WARNING RE-//
+T6.. REDUCTION LEVELS, CHAFF DISENSE (BASELINE) ARE ZERO, INDICATING.//
+T6.. CEIVER. THE DEFALT VALUE ARE (INCLUDED).//
+T6.. NCNE OF THESE FEATURES ARE INCLUDED. VARY WITH THE THREE.//
+T6.. VULNERABILITY REDUCTION FEATURES. VARY WITH THE THREE.//
+T6.. TYPES OF AIRCRAFT. SELECT THOSE FEATURES OF 1 (BASELINE).//
+T6.. DESCRIBE YOUR IMPROVEMENTS. MINIMUM VALUES OF 1 (BASELINE).//
+T6.. INDICATE ENTER ANY INTEGER TO RETURN TO MENU 2
+T6.. READ(5,*)IJK
GO CALL FRTCMS('CLRSCRN ')
9973 WRITE(4,7973)
7973 FORMAT(4,*)
+T6.. T6.. COMBAT SCENARIO SECTION IS DIVIDED INTO TWO SUBSECTIONS.//
+T6.. IN THE COMBAT SCENARIO SECTION VALUES ARE ENTERED TO SPECIFICALLY//
+T6.. BY THE SELECTION OF AIRCRAFTS. THE MISSION PARAMETERS ARE DEFINI//
+T6.. THAT MIGHT BE CONSIDERED AS TARGETS. THESE INCLUDE.//
+T6.. IN THREAT SELECTION, THE THREAT DENSIITIES AND THREAT.//
+T6.. DIAMETERS ARE ENTERED. THE THREATS FOR THE LONG RANGE STRIKE
C- #2
+T6.. MISSION ARE: AIR-TO-AIR IR MISSILES AND HIGH ALTITUDE SAM.//
+T6.. ENTER ANY INTEGER TO RETURN TO MENU 3
+T6.. READ(5,*)IJK
GO CALL FRTCMS('CLRSCRN ')
9974 WRITE(4,7974)
7974 FORMAT(4,*)
+T6.. THE SUSCEPTIBILITY ASSESSMENT SECTION HAS TWO SUBSECTIONS.//
+T6.. OF THE AIRCRAFT FROM THE THREAT TO THE A/C AT CPA.//
+T6.. AND THE AIRCRAFT FROM THE THREAT TO THE A/C AT CPA.//
+T6.. THAT IS THE SAME AS THE AIRCRAFT FROM THE THREAT TO THE A/C AT CPA.//
+T6.. AS THE ALTITUDE IS 1.41 HIGHER. THE ALTITUDE SEPARATELY FOR EACH.//
+T6.. CPA THAT PROBABILITY OF HIT IS DEFINED. THIS FORM IS CONSISTENT, WHERE.//
+T6.. AIRCRAFT TO THE THREAT. PROBABILITY THAT A NON-MANEUVERING A/C.//
+T6.. WOULD BE HIT BY THE THREAT IS P(H). THIS IS THE MANEUVER FACTOR.//
+T6.. AND F(C) IS P(H) * F(C). ENTER ANY INTEGER TO RETURN TO MENU 4
+T6.. READ(5,*)IJK

```









```

GO TO 598
*****
EXIT
*****
C- #3
-----
CONTINUE
*****
DETERMINE TCGW *****
CALL ESRWT(ES,EC,EL,TW,WS,WT,
IFS,IFV,JAM,IRCS,XMA,XMD,XMM,IWARN,ICHAFF,
BLTOGW,TOGW)
*****
C- #3
-----
** TC SAVE DATA *****
REWINC 2
WRITE(?,IO12)TW,WS,WT,EC,EL,ES,JAM,IRCS,IWARN,ICHAFF,IFS,
IFV,XMA,XMD,DSR,SAMH,AAH,AAD,
PCAR,PLSM,PHAR,PHSM,AVAA,PKHAA,PKHSM,PSAR,
PSSM,ACR,XNPASS,ACRI,NSR,XNPASS,NS,
ACR2,TCISR,TTACK,TOTACL,TOTACR,BLTOGW,TOGW
*****
C- #3
-----
FRTCMS('CLRSCRN ')
CALL (4,9200)TW,WS,WT,XMD,AAH,XMA,AAD,XMM,SAMH,SAMD
WRITE (4,2220)JAM,IFS,IRCS,IFV,IWARN,ICHAFF
WRITE (4,9201)PSAR,PCAR,PHAR,PKHAA
WRITE (4,9202)PSSM,PCSM,PHSM,PKHSM
WRITE (4,9203)
WRITE (4,9204)
WRITE (4,9205)
WRITE (4,9206)ACR1,NSPT,XNPASS,NS,ACR2,TOTSR,TOTACK,TOTACL,TOTACR
WRITE (4,9207)BLTOGW,TOGW
WRITE (4,8282)
WRITE (4,8282)
GO TO 8
*****
CONTINUE
CALL FRTCMS('CLRSCRN ')
WRITE(4,9999)
FORMAT('0',) TO PRINT YOUR RESULTS AND EXIT ENTER "0"/
+ ' TO PRINT YOUR RESULTS AND REENTER PRCGRAN ENTER "1"/
+ ' TO EXIT WITHOUT A PRINT ENTER "2"/
PEAD(5,'#JJK
IF(IJK.FQ.2) GO TO 9999
*****
C- #4
-----
WRITE (6,9200)TW,WS,WT,XMD,AAH,XMA,AAD,XMM,SAMH,SAMD
WRITE (6,2220)JAM,IFS,IRCS,IFV,IWARN,ICHAFF

```

```

WRITE (6,9201) PSAR,PCAR,PHAR,PKHAA
WRITE (6,9202) PSSM,PDSM,PHSM,PKHSM
WRITE (6,9203) ACRI,NSPT,XNPASS,NS,ACR2,TOTSR,TOTACK,TOTACL,TGTACR
WRITE (6,9204) BLTOGW,TOGW
IF(IJK.EQ.1) GO TO 1022
C- #3, #4
9200 * * * * * ** LCNG RANGE STRIKE AIRCRAFT * * * * *
130,125 * * * * * PERFORMANCE FEAT.,T31,F8.2,
142, * * * * * THRUST WEIGHT,T31,F8.2,
* * * * * WING LCADING,T31,F8.2,
* * * * * ORDNANCE WEIGHTS,T31,F8.2,
* * * * * * * * * * THREAT PARAMETERS * * * * *
142, * * * * * PENETRATION DISTANCE,T31,F8.2,
* * * * * A/A DENSITY,T65,F8.2,
142, * * * * * PENETRATION ALTITUDE,T31,F8.2,
142, * * * * * A/A CLAMPETER,T65,F8.2,
142, * * * * * PENETRATION MACH NR,T31,F8.2,
142, * * * * * SAM DENSITY,T65,F8.2,
142, * * * * * SAM DIAMETER,T65,F8.2,
C- #3, #4
9201 * * * * * ** SURVIVABILITY ASSESSMENT * * * * *
18, * * * * * P(S) P(H) P(K/H)
9202 * * * * * VS A/A MISSILE ,4(F4-2,5X)
9204 * * * * * VS HIGH ALT SAM ,4(F4-2,5X)
9205 * * * * * INITIAL A/C ,F8.0, * * * * * NUMBER OF RAIDS ,17,
9206 * * * * * PASSES/SORTIE ,F8.0, * * * * * SORTIES FOR REPAIR ,F8.0,
* * * * * A/C READY ,F8.0, * * * * * TOTAL SORTIES ,F8.0,
* * * * * TCTAL TARGETS ,F8.0, * * * * * TCTAL A/C LDST
9220 * * * * * ** BASELINE TOGW ,F10.2,T36,ENHANCED TOGW,F10.2)
8282 * * * * *
9207 * * * * * SUBROUTINE SSRPDA(JAMS,IRCSS,PDCSI)
* * * * * SUBROUTINE SSRPDS(JAMMER,RCS,DSR,PCF)
* * * * * REAL PDCM(10)
* * * * * DATA PDCM/99,986,.985,.98,.96,.93,.91,.87,.805,.76/
* * * * * PDCS=PDCM(IRCSS+1)
* * * * * IF(JAMS.NE.0) PDCS=PDCM(10)
* * * * * RETURN
* * * * * END
* * * * * SUBROUTINE SSRPDS(JAMMER,RCS,DSR,PCF)
* * * * * INTEGER A,B,FCS,JAMMER,Q

```

```

DIMENSION H(2,9,6),F(101),PDI(101),XX(101)
DATA H/37.,10.,36.,10.,31.,9.,29.,8.,26.,8.,22.,5.6.,21.,6.,17.,5.5.,
A13.,4.,21.,5.5.,17.,5.,15.,2.,4.,2.,11.,3.,8.,8.,2.,7.,5.,1.9.,5.5.,1.8.,0.,
B23.,5.,7.,21.,5.5.,17.,5.,15.,2.,4.,2.,11.,3.,8.,8.,2.,7.,5.,1.9.,5.5.,1.8.,0.,
C1.,8.,15.,5.,4.,8.,12.,3.,7.,10.,5.,3.,2.,7.,9.,2.,5.,6.,1.,9.,4.,8.,1.,1.,3.,5.,
C16.,8.,3.,1.,8.,3.,2.,4.,7.,5.,2.,2.,5.,6.,1.,7.,4.,2.,1.,3.,3.,5.,1.,0.,1.,0.,1.,
C10.,8.,3.,1.,4.,5.,2.,1.,4.,7.,1.,4.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,
D16.,3.,1.,4.,3.,4.,1.,1.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,
E14.,3.,1.,3.,1.,2.,4.,1.,1.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,
F1.,4.,3.,1.,3.,1.,2.,4.,1.,1.,0.,1.,0.,1.,0.,1.,0.,1.,0.,1.,
C***** I=RCS+1
C***** J=JAMMER+1
C***** X=H(1,I,J)
C***** S=H(2,I,J)
C-----
C- #1 CON1=1./((S**CRT(2.*3.14159))
CON2=-5/S**2
C***** XI=X+4*S
C***** STEP=-S/12.5
C***** F(1)=0.9
C***** XX(1)=XI
PDF=0.
C***** DO 10 JJ=1,100
C***** CCN1=EXP(CCN2*(XI-X)**2)
F(JJ+1)=STEP*(F(JJ+1)+F(JJ))
AREA=-.5*STEP*(F(JJ+1)+F(JJ))
PDI(JJ+1)=PDF
IF (X).LT.DSRIGO TO 5
PDF=PD
CONTINUE
XX(JJ+1)=XI
PD=PD+AREA
C***** CONTINUE
C***** IF(PDF.GT.0.01) GC TC 20
PDF=100
CONTINUE
END
SUBROUTINE SSRPHR(WS,XMM,XMA,IWARN,ICHAFF,DSR,PDAM)

```

```

*****
***** A/A RADAR MISSILE P(F) *****
***** WSS = WS/100 *****
***** XMR = XMA/10000 *****
***** XXMR = DSR *****
***** PH = 1.0 *****
***** CALL SRFC (ICHAFF,FC) *****
***** MCDIFIED FOR CHAFF *****
***** PH = PH * FC *****
***** CALL SRFA (XXMR, IWRN, FA) *****
***** MCDIFIED FOR MANUVERING *****
***** WSS**2 = .08246 * XMS * XMM *****
***** FM = 1. + .961 * WSS**2 - .08246 * XMS * XMM *****
***** IF (FM.LT.0.01) FM = 0.01 *****
***** IF (FM.GT.1.0) FM = .99 *****
***** XMF = 1. - (1. - FM) * FA *****
***** PDAM = PH1 * XMF *****
***** RETURN *****
END
SUBROUTINE SSRPHS (WS, XMM, XMA, IWRN, ICHAFF, DSR, PDAM)
WSS=WS/100.
XMS=XMA/10000.
XXMR = DSR
***** PH = 1.0 *****
***** CALL SRFC (ICHAFF,FC) *****
***** MCDIFIED FOR CHAFF *****
***** PH = PH * FC *****
***** CALL SRFA (XXMR, IWRN, FA) *****
***** MCDIFIED FOR MANUVERING *****
***** WSS**2 + .165674 * WSS * XMS *****
***** FM = 1. - .25393 * XMM**2 + .165674 * WSS * XMS *****
***** IF (FM.LT.0.01) FM = 0.01 *****
***** IF (FM.GT.1.00) FM = 1.00 *****
***** XMF = 1. - (1. - FM) * FA *****
***** PDAM = PH1 * XMF *****
***** RETURN *****
END
SUBROUTINE SRFA (XXMR, IWRN, FAS)
***** ALERTRON FACTCR *****
***** REAL DATA FESM/1. *****
***** DATA FESM/0.1, .26, .40, .50, .60, .70, .80, .90, .100, .110, .120, .130, .140, .150, .160, .170, .180, .190, .200, .210, .220, .230, .240, .250, .260, .270, .280, .290, .300, .310, .320, .330, .340, .350, .360, .370, .380, .390, .400, .410, .420, .430, .440, .450, .460, .470, .480, .490, .500, .510, .520, .530, .540, .550, .560, .570, .580, .590, .600, .610, .620, .630, .640, .650, .660, .670, .680, .690, .700, .710, .720, .730, .740, .750, .760, .770, .780, .790, .800, .810, .820, .830, .840, .850, .860, .870, .880, .890, .900, .910, .920, .930, .940, .950, .960, .970, .980, .990, 1.0 *****
***** DATA PRM/0.1, .1, .2, .3, .4, .5, .6, .7, .8, .9, 1.0 *****
***** DATA FVM/1.0, .995, .95, .943, .9, .868, .85, .8, .75, .712, .7, .65, .6 *****

```

```

A 1.55,.5,.45,.4,.357,.25,.3,.25,.216,.2,.165,.15,.13,.114,.101/
IF(IWARN.EQ.1)GO TC 15
KEY=1
CONTINUE
IF(MRM(KEY).LT.XMRS)GO TO 5
DELTAY=MRM(KEY)-MRM(KEY-1)
FVS=((XMRS-MRM(KEY-1))/DELTAY)*DELTAY+FVM(KEY-1)
GO TO 20
CONTINUE
J=1
CONTINUE
J=J+1
IF(XMRS.GE.MRSM(J))GO TO 10
DX=MRSM(J-1)-MRSM(J)
DY=(FESM(J-1)-FESM(J))/DX*CY+FESM(J-1)
FAS=FES
CONTINUE
RENC
C***** CHAFF FACTOR *****
C***** SUBROUTINE SRFC(ICAFF,FC) *****
C***** REAL PBTSM(17) *****
C***** DATA PBTSM/ .00,.19,.35,.49,.6,.68,.74,.8,.83,.86,.9,.92,.935, *****
C***** .95,.96,.97,.98/ *****
A
PBTSM=0.
IF(ICHAFF.EQ.0)GO TO 19
NBUNDS=4
PBTSM=(NBUNDS+1)
FC=1.
RETURN
ENC
C***** VULNERABLE AREA AND P(K/H) VS A/A RADAR MISSILE *****
C***** SUBROUTINE SSPAVA(IFS,IFV,AV,UPKHS) *****
F1=0.
F2=0.
F3=0.
F4=0.
IF (IFS.EQ.2) F1 = 1.
IF (IFS.EQ.3) F2 = 1.

```





```

C***** F4 = 0. CANNOT BE AN INDEX *****
      IY=JAM+1
      IX=IRCS+1
      J=0
      CONTINUE
15      J=J+1
      IF (CEPS(J)) GO TO 15
      IF (GRAPH PCINT EXTRACT) *****
C*****      DX=CEPSM(J-1)-CEPSM(J) *****
      DY=CEPSM(J-1)/CEPSM(J) *****
      CEPS=(CEPS-CEPSM(J-1))/DX+DY+CEPSM(J-1) *****
C*****      CEPS=VALUE OF CEP IS 1000 FT *****
      IF (MAX PS GT 1000.1) CEPS=1000. *****
      IF (IFS.EQ.3) F1 = 1. *****
      IF (IFS.EQ.3) F2 = 1. *****
      IF (IFS.NE.4) GC TC 5 *****
      IF (F1 = 1. *****
      IF (F2 *****
      IF (F3 *****
      IF (F4 *****
5      CONTINUE *****
      IF ((IFV.EQ.2).CR.(IFV.EQ.3)) F2 = 1. *****
      IF ((IFV.EQ.4).CR.(IFV.EQ.5)) F4 = 1. *****
      IF (F1 *****
      IF (F2 *****
      IF (F3 *****
      IF (F4 *****
10     CONTINUE *****
      TOT = F1 + F2 + F3 + F4 *****
      IF (TOT.GT.3.5) GO TC 4 *****
      IF (TOT.GT.2.5) GO TC 3 *****
      IF (TOT.GT.1.5) GO TC 2 *****
      IF (TOT.GT.0.5) GO TC 1 *****
      PKHS=1 *****
      GO TO 20 *****
1      P1=(.99857-.000525606*CEPS+.000000341377*CEPS**2)*F1 *****
      P2=(.99851-.0000073615*CEPS-.00000042731*CEPS**2)*F2 *****
      P3=(.98418-.000077649*CEPS-.00000152060*CEPS**2)*F3 *****
      P4=(.98867-.000114141*CEPS+.000000115539*CEPS**2)*F4 *****
      PKHS=F1+P2+P3+P4 *****
      GO TO 20 *****
2      P1=(.99857-.000127961*CEPS+.000000134138*CEPS**2)*F1 *****
      P2=(.98914-.0000836242*CEPS+.000000365215*CEPS**2)*F2 *****
      P3=(.97357-.000117606*CEPS-.00000149176*CEPS**2)*F3 *****
      P4=(.95001-.0000246527*CEPS-.0000000471454*CEPS**2)*F4 *****
      P5=(.98489-.000238044*CEPS+.000000126457*CEPS**2)*F4 *****
      P6=(.96155-.0000721483*CEPS-.0000000388464*CEPS**2)*F4 *****
      PKHS=F1+P2+P3+P4+P5+P6 *****
      GO TO 20 *****

```



```

H4 = ACCVER * PH1
XK4 = H4 * PKH1
A4 = F4 - XK4
*****THREAT 2*****
H5 = (ACCVER-H4) * PF2
XK5 = F5 * PKH2
A5 = H5 - XK5
*****THREAT 3*****
C- #1 -----
H6 = (ACCVER-H4-H5) * PH3
XK6 = H6 * PKH3
A6 = F6 - XK6
*****TCTALS FCR SORTIE*****
ACNFT = ACRI-F1-H2-H3-F4-H5-H6
ACDAM = A1 + A2 + A3 + A4 + A5 + A6
ACKIL = XK1 + XK2 + XK3 + XK4 + XK5 + XK6
*****FOR NEXT SORTIE*****
ACR2 = ACNHT
TOTSR = ACRI
TOTACK = ATAC
TOTACL = ACKIL
TCTACR = ACDAM

RETURN
ENC
SUBROUTINE CAMP (
&XL1,XH1,D1,PKH1,PS1, XL2,XH2,D2,PKH2,PS2, XL3,XH3,D3,PKH3,PS3,
&ACRI,NSRT,XNPAS,NS, ACR2,TCISR,TOTACK,TOTACL,TCTACR)
*****
*****XL-PENDIS F-THREAT DENSITY D- THREAT DIAMETER*****
*****
*****DIMENSION ACR(200)*****
TOTSR = 0.
TCTACK = 0.
TCTACL = 0.
TCTACR = 0.
C
C
C
C
W1 = XL1 * XH1 * D1 / 100.
W2 = XL2 * XH2 * D2 / 100.
W3 = XL3 * XH3 * D3 / 100.
PSM1 = PS1 * W1
PSM2 = PS2 * W2
PSM3 = PS3 * W3
C
PH1 = (1. - PSM1) / PKH1
PH2 = (1. - PSM2) / PKH2
PH3 = (1. - PSM3) / PKH3

```

```

C
  ACR(I) = ACR1
  DO 10 I = 1, NSRT
  ***** INGRESS *****
  H1 = ACR(I) * PH1
  XK1 = H1 - XK1
  A1 = H1 - XK1
  H2 = (ACR(I) - H1) * PH2
  XK2 = H2 - XK2
  A2 = H2 - XK2
  H3 = (ACR(I) - H1 - H2) * PH3
  XK3 = H3 - XK3
  A3 = H3 - XK3
  ***** OVER TARGET *****
  ACOVER = ACR(I) - H1 - H2 - H3
  ATAC = ACOVER * XNPAS
  ***** EGRESS *****
  H4 = ACOVER * PH1
  XK4 = H4 - XK4
  A4 = H4 - XK4
  H5 = (ACCOVER - H4) * PF2
  XK5 = H5 - XK5
  A5 = H5 - XK5
  -----
  H6 = (ACCOVER - H4 - H5) * PH3
  XK6 = H6 - XK6
  A6 = H6 - XK6
  ***** TCIALS FCR SCRTIE *****
  ACNHT = ACR(I) - H1 - H2 - H3 - H4 - H5 - H6
  ACDAM = A1 + A2 + A3 + A4 + A5 + A6
  ACKIL = XK1 + XK2 + XK3 + XK4 + XK5 + XK6
  ***** FOR NEXT SORTIE *****
  TICR = TICR + ACDAM
  TICRACR = TICRACR / FLGAT(NS)
  TICRACR = TICRACR - ACROUT
  ACR(I+1) = ACNHT + ACROUT
  TICR = TICR + ACR(I)
  TICRACK = TICRACK + ATAC
  TICRACK = TICRACK + ACKIL
  10 CONTINUE
  RETURN
  ENCL
  SUBROUTINE ESRWT(ES, EC, EL, TW, WS, WT,
  *****
  ***** DETERMINATION SUBROUTINE *****
  ***** TCGW *****
  ***** @ IFS, IFV, JAM, IRCS, PENALTY, PENDIS, PENMAC, IWARN, ICHAFF,

```

BLTCGW,TCGW)  
REAL A,B,C,D,E,F,G,H,I,J  
TCGW = C.O.  
BLTCGW = 0.  
A = TW  
B = WS  
C = PENALT  
D = PENMAC  
E = PENDIS  
F = WT  
G = W.O.  
H = O.  
I = O.  
J = O.

a

```
C*****  
C- #1  BLTCGW ***** RLTCGW *****  
      +. 95528.1-2.42453*C-163690.*G+5771.94*A*D+1.00566*A*H  
      -. 00404312*C+1.41229*B*E-13033.3*B*G+.00023164*C*C  
      +. 388375*C*F+1611310*D*G+.872068*C*I+138.061*D*E  
      +. 1.30706*D*F+1611310*D*G+.411576*D*H+37635.65*D*J+5560.39*E*G  
      -. 1827710.*G*I+2504410.*G*J  
      A/C TOGW OF DESIGN WITH SURVIVABILITY ENHANCEMENT  
      THE FOLLOWING ASSUMPTION MADE; 23 MM  
      ALL SELF-SEALING TANKS; HAVE EQUAL VOLUME OF SINGLE SUMP TANK(S)  
      TWO SUMP TANKS; HAVE EQUAL VOLUME OF  
      DUAL SUMP TANKS; HAVE EQUAL VOLUME OF  
      INTERNAL FOAM VICE ULLAGE INERTING  
      FIRE EXTINGUISHING VICE VOID FOAM  
      IF (JAM.EQ.C) GC TC 40  
      SREF=BLTCGW/WS  
C- #1  T1=1.7/1.+0.1296**PENMAC**2)***.648  
      T2=1.8534/PENMAC**2  
      T3=1.793-1.036/(PENMAC**2)+.162/(PENMAC**4)  
      G = (0.0536*T1+T2+T3**PENMAC**5)/SREF  
40 *****  
C ***** WEIGHT INCREASE CALCULATIONS *****  
C ***** FR = FUEL REQUIRED FOR MISSION *****  
C- #1  FR=9499.85-.772933*C+24414.*D-464065.*A*G+.414073*A*H  
      *  
      -.C0681604*8*F
```

```

a -10611.*8*G+345.404*8*I+.0000173927*C*-545816*C*D-.0C11389*C*E
a +.31419*C*I+69.4257*D*E+.68633*D*F+944143.*D*G+.0876982*D*H
a -19218.*8*C*J+2979.94*E*G-23.2469*F*I+10929000.*G*G+20.0457*G*H
a -1573550.*G*I+1517220.*G*J+.C00017662*H*H
C*****WEIGHT INCREASE DUE TO SELF-SEALING *****
XNT = 2. EQ.1) OR. (IFS.EQ.2)) XNT = 0.
IF (IFS.EQ.3) XNT = 1.
IF (IFS.EQ.2) XNT = 1.45*(1.75*(FR/6.6))*.64*XNT*.11
C*****WEIGHT INCREASE DUE TO INTERNAL FOAM *****
WF = 0.
IF (IFV.EQ.2) CR. (IFV.EQ.6)) WF = .C186 * FR/6.6
C*****WEIGHT INCREASE DUE TO FIRE EXTINGUISHING *****
WFE = 0.
XV = 4.72 * ( EC + ES) * EC * EL
IF (IFV.EQ.5) OR. (IFV.EQ.6)) WFE = 10.5 * XV*.26
C*****WEIGHT INCREASE DUE TO ULLAGE INERTING *****
WIRT = .015 * (FR/6.6)*.92 * XNT
C*****WEIGHT INCREASE DUE TO EXTERNAL FOAM *****
WEF = 0.
IF (IFV.EQ.4) WEF = 2.65*(BLTGM-FR)*.001
C*****WEIGHT INCREASE DUE TO RAM *****
XS = 0.
IF (IRCS.EQ.1) XS = 10.
IF (IRCS.EQ.2) OR. (IRCS.EQ.3)) XS = 50.
IF (IRCS.EQ.4) XS = 60.
IF (IRCS.EQ.5) XS = 70.
IF (IRCS.EQ.6) XS = 80.
IF (IRCS.EQ.7) OR. (IRCS.EQ.8)) XS = 80. + BLTGM/WS *.63
C*****WEIGHT INCREASE DUE TO RWR *****
WWRAM = 0.
IF (IWARN.EQ.1) WWRAM = 50.
C*****WEIGHT INCREASE DUE TO RADAR JAMMER *****
WJW = 0.
IF (JAM.EQ.1) WJW = 80.
IF (JAM.EQ.2) WJW = 100.
IF (JAM.EQ.3) WJW = 200.
IF (JAM.EQ.4) WJW = 500.
IF (JAM.EQ.5) WJW = 1000.
C*****WEIGHT INCREASE DUE TO CHAFF DISPENSER *****
WCD = 0.
IF (ICHAFF.EQ.1) WCD = 86.
C*****WEIGHT INCREASE DUE TO SUBMERGED STORE *****
WSOR = 0.
IF (IRCS.EQ.7) OR. (IRCS.EQ.8)) WSCR = 1.13 * WT/100.
C*****

```

```

C***** TOTAL WEIGHT INCREASE *****
H = WSSP+WF+WE+WIRT+WEF+WRAM+NEW+WJW+WCD+WSOR
C*****
C***** TOTAL TQGN OF ENHANCED A/C*****
C*****
C- #1 -----
          TOGW = 95528.1-2.43453*C-163669C.*G+5771.94*A*D+1.00566*A*H
          +.00404312*8*C-1.41229*8*E-130333.3*8*G+.00023164*C*C
          -.388375*C*D+.C000132684*C*H+.872068*C*I+138.061*D*E
          +1.30706*D*F+1611310.*D*G+.411576*D*H+37639.65*D*J+5560.39*E*G
          -22.6367*F*I-4.18043*F*J+13373400.*G*G+41.0552*G*H
          -1827710.*G*I+2504410.*G*J
          RETURN
          END

```

```

*****
C*** STRFLT < STRIKE FLCT >
C*** PROGRAM TO GRAPH P(K)'S VS. THREATS AGAINST
C*** LCNG RANGE STRIKE AIRCRAFT
C***
*****
C*** DIMENSION YC(3),Y1(2),Y2(2),Y3(2),XC(2),IPKRAY(400)
C*** * LABEL(12)
C*** * DATA LABEL,SAM,,,,,A/A,,MISS,,ILE ,
C*** DATA YC/0.3,0.7
C*** DATA XC/1.5,2.57
C
C READ(8,800)(Y1(I),I=1,2),(Y2(I),I=1,2),(Y3(I),I=1,2)
C FORMAT(F4.2)
C
TEK618
CALL FWRDT('MOVIE')
CALL FLOWUP(1,4)
CALL PAGE(8,5,11.)
CALL NCBRDR(6,8.5)
CALL AREA2D('THREATS',100)
CALL XNAME('P(K)')$,100)
CALL SWISSM(50,1,003,1)
CALL SHDCHR('STAND')
CALL RASALF('L/CSTC')
CALL MIXALF('LONG RANGE STRIKE AIRCRAFT',100,1,5,2)
CALL HEADIN('L(CSS) R(ATE) VS. T(HREAT) T(TYPE)$',100,1,1,2)
CALL YPEVTK
CALL YAXANG(C.)
CALL XTICKS(0)
CALL XAXEND('NCFENDS')
CALL YAXEND('NOLAST')
CALL SWISSL
CALL HEIGHT(.12)
CALL FRAME
CALL THKERM(.10)
CALL XLABGR('LABEL,3,4,0,,1,1,1)
CALL FLREC('C,01,8,14,29,0,6,0)
CALL BLREC(4,3,7,7,1,7,1,0,0)
CALL BLKEY(IC)
CALL CLUSTR(3,.08)
CALL HFLGHT(.08)
CALL RAPDCC('SECCND', 'OUTSIDE',2)
CALL BLBAR

```



```

CALL VBAR$ (. LABEL, YO, Y1, 2)
CALL VBAR$ (. LABEL, YO, Y2, 2)
CALL VBAR$ (. LABEL, YO, Y3, 2)
CALL HEIGHT (.05)
CALL CUT
CALL GRID (0, 2)
CALL RESET (DOT, 1)
CALL HEIGHT (.10)
CALL FLOFF (IC)
CALL IN-LINE$ (IPKRAY, 40, 40)
MAXL IN-LINE$ (A$ELINE) $, 1, IPKRAY, 1)
CALL LINES (. 1ST C (ESIGN) $, ., IPKRAY, 2)
CALL LINES (. 2ND C (ESIGN) $, ., IPKRAY, 3)
CALL LEGEND (IPKRAY, 3, 4, 5, 7, 6)
CALL ENDP (C)
CALL DCNEPL
CALL STOP
END

```



```

C- #5  DATA K5/'KH','KD'//
        DATA K6/'AR','AO','LS','SS','CE'//
        DATA KK/'HP','TN','EX','RT'//
        DATA JJ/'Y','N'//
C- #3  ***** TO SAVE DATA *****
C *****
CALL FRTCMS('CLRSCRN ')
WRITE(4,1010)
FORMAT(1,1)
* T6, IF THIS IS YOUR FIRST TIME THROUGH SUPPORT OR YOU WISH //
* T6, TO USE THE DEFAULT VALUES/PARAMETERS ENTER...0 //
* T6, TO USE DATA SAVED FROM YOUR LAST RUN ENTER...1 //
* T6, WARNING //
* T6, --CO NOT ENTER 1 IF THIS IS YOUR FIRST RUN OR IF YOU HAVE //
* T6, ERASED YOUR SUPPORT DATA FILE FROM YOUR DISK-- //
C READ(4,1011)
FORMAT(1,1)
IF(11.EQ.0)CC TO 1021
IF(11.EQ.1)CC TO 1022
CONTINUE
REWRITE(4,1012)
* * *
* * *
* * *
FORMAT(1,1)
GO TO 1021
CONTINUE
DATA T1/.55/,WS/90./,WT/8000./,B/4C./,XL/40./,W/4./,EC/1.5/,ED/3.5
& /,EL/12./
DATA JAM/0/,IRCS/0/,IWARN/0/,ICHAFF/0/
DATA YFS/1/,IFV/1/,FE/1/,TEA/1/,TEP/1/,ICS/1/,ICA/1/
DATA XMR/15C./,XMA/10CC./,XMT/60./
DATA AAH/1/,AAAD/3./,SAM/9588/
DATA PCAR/1./,PCAR1/1./,PDSM/9588/
DATA PHR/9007./,PHG/1416./,PHSM/VAS/1215/
DATA VAAA/100./,PKHAAA/4531/,NS/100./,PKHSM/1./
DATA PSAR/5919/,PSAC/9358/,PESM/382/
C- #3  DATA ACR/10C./,XINPAS/1./
        DATA ACR1/100./,NSRT/20/,XNPASS/1./,NS/4/
        DATA ACR2/29.50/,TCTSR/1103.24/,TOTACK/1038.56/,TOTACL/64.51/

```

```

C- #5 DATA TCTACR/5.9%/ ,BLTCCG/28945.09/ ,TOGW/28945.09/
C*****
C MAIN MENU DISPLAY *****
C*****
100 CCNT INUE *****
1 CALL FRTGMS('CLRSCRN ') *****
WRITE(4,1001) *****
FORMAT(1,1) *****
+T6, 'FCR AN EXPLANATION MENU (1) SELECT A CODE AS FOLLOWS: '// *****
+T6, 'AIRCRFT DESIGN SELECTION *****
+T6, 'ACCIDENT SCENARIO SELECTION *****
+T6, 'SUSCEPTIBILITY ASSESSMENT *****
+T6, 'VULNERABILITY ASSESSMENT *****
+T6, 'TC TRANSFER TO OTHER MENUS *****
+T6, 'TC EXIT (CR) *****
FORMAT(5,2000) K1C *****
READM(1,4) *****
IF(K1C.EQ.K1(1)) GC TC 110 *****
IF(K1C.EQ.K1(2)) GC TC 120 *****
IF(K1C.EQ.K1(3)) GC TC 130 *****
IF(K1C.EQ.K1(4)) GC TC 140 *****
IF(K1C.EQ.K1(5)) GC TC 150 *****
IF(K1C.EQ.K1(1)) GC TC 9971 *****
IF(K1C.EQ.K1(2)) GC TC 998 *****
C- #3 IF(K1C.EQ.K1(3)) GC TC 1061 *****
WRITE(4,1200) *****
FORMAT(1) *****
GO TO *****
C*****
C MENU 2 *****
C*****
110 CALL FRTGMS('CLRSCRN ') *****
CCNT INUE *****
WRITE(4,110) *****
FORMAT(1,1) *****
+T6, 'FOR AN EXPLANATION MENU (2) DESIGN, ENTER A CODE AS FOLLOWS: '// *****
+T6, 'A/C PERFORMANCE INDICATORS *****
+T6, 'SUSCEPTIBILITY FEATURES *****
+T6, 'VULNERABILITY FEATURES *****
+T6, 'TC TRANSFER TO OTHER MENUS *****
+T6, 'TC TRANSFER TO OTHER MENUS *****
IF(K2C.EQ.K2(1)) GC TC 210 *****

```

```

-----
IF(K2C.EQ.K2(2)) GC TC 220
IF(K2C.EQ.K2(3)) GO TC 230
IF(K2C.EQ.KK(1)) GC TC 9972
IF(K2C.EQ.KK(2)) GO TC 998
C- #3 IF(K2Q.EQ.KK(4)) GO TC 7
      WRITE(4,120C)
      GO TO 2
      *****
      MENU 3 COMBAT SCENARIO
      *****
      CALL FRTCMS('CLRSCRN ')
      CCNTINUE
      WRITE(4,112C)
      FORMAT(1,1, MENU (3) COMBAT SCENARIO, ENTER A CODE AS FOLLOWS:)//
      +T6, 'FOR AN EXPLANATION',
      +T6, 'MISSION PROFILE',
      +T6, 'THREAT SELECTION',
      +T6, 'TC RETURN TO MENU (1)',
      +T6, 'TC TRANSFER TO OTHER MENUS',
      READ(5,20C) K3C
      IF(K3C.EQ.K3(1)) GO TC 310
      IF(K3C.EQ.K3(2)) GO TC 320
      IF(K3C.EQ.KK(1)) GO TC 9973
      IF(K3C.EQ.KK(2)) GC TC 998
C- #3 IF(K3C.EQ.KK(4)) GC TC 7
      WRITE(4,120C)
      GO TO 3
      *****
      MENU 4 SUSCEPTIBILITY ASSESSMENT
      *****
      CALL FRTCMS('CLRSCRN ')
      CCNTINUE
      WRITE(4,112C)
      FORMAT(1,1, MENU (4) SUSCEPTIBILITY ASSESSMENT,)//
      +T6, 'ENTER AN EXPLANATION',
      +T6, 'PROBABILITY OF DETECTION',
      +T6, 'PROBABILITY OF HIT',
      +T6, 'TC RETURN TO MENU (1)',
      +T6, 'TC TRANSFER TO OTHER MENUS',
      READ(5,20C) K4C
      IF(K4C.EQ.K4(1)) GO TC 410
      IF(K4C.EQ.K4(2)) GO TC 420
      IF(K4C.EQ.KK(1)) GO TC 9974
      IF(K4C.EQ.KK(2)) GC TC 998
-----
C- #3

```













```

+T6, ENTER 0 FCR NO CHANGE REQUIRED')
READ(5,1211)I1
IF(I1.EQ.0)GO TC 110
GO TO (231,232,233,234,235,236,237),I1
WRITE(4,120C)
GD
231 CCNTINUE
CALL FRTCMS('CLRSCRN ')
WRITE(4,1231)
231 FORMAT
+T6, 1 FUEL SYSTEM, GENERAL
+T6, 2 TANK, NO SELF-SEALING
+T6, 3 SINGLE SUMP TANKS, WITH SELF-SEALING
+T6, 4 DUAL SUMP TANKS, NO SELF-SEALING
+T6, 5 DUAL SUMP TANKS, WITH SELF-SEALING
+T6, 6 DUAL SUMP TANKS, EXTRA SELF-SEALING
+T6, 7 DUAL SUMP TANKS, NO SELF-SEALING
+T6, 8 DUAL SUMP TANKS, WITH SELF-SEALING
+T6, 9 ENTER THE PROTECTION NUMBER IN I1 FORMAT')
IFV=I1
GO TO 230
CCNTINUE
CALL FRTCMS('CLRSCRN ')
WRITE(4,1232)
232 FORMAT
+T6, 1 TANKS ADJACENT TO DRY BAYS, HIGH SURFACE TEMP.
+T6, 2 TANKS ADJACENT TO DRY BAYS, WITH ELECTRICAL EQUIPMENT
+T6, 3 TANKS ADJACENT TO DRY BAYS WITH INERTING EQUIPMENT
+T6, 4 INTERNAL FOAM OR INERTING FOR TANK ULLAGES
+T6, 5 INTERNAL FOAM OR FIRE EXTINGUISHING FOR VOIDS
+T6, 6 BOTH INTERNAL AND EXTERNAL PROTECTION 2 OR 4 & 3 OR 5
+T6, 7 ENTER THE DESIRED PROTECTION LEVEL IN I1 FORMAT')
IFV=I1
GO TO 230
CCNTINUE
CALL FRTCMS('CLRSCRN ')
WRITE(4,1233)
233 FORMAT
+T6, 1 FUEL/ENGINE INTERFACE
+T6, 2 FUEL AROUND ENGINE WITH PROTECTION
+T6, 3 FUEL AROUND DUCTS
+T6, 4 FUEL AROUND DUCTS WITH PROTECTION
+T6, 5 DUCTS AROUND FUEL
+T6, 6 DUCTS AROUND FUEL WITH PROTECTION
+T6, 7 FUEL FORWARD OF ENGINE
+T6, 8 POSITIVE FUEL/ENGINE SEPARATION

```

```

+T6, ENTER THE PROTECTION NUMBER IN I1 FCRMAT'
  READ(5,1211)I1
  IFC=I1 230
  GCNT INDE
  CALL FRTCMS(1,CLRSCRN *)
  WRIT(1,1) ENGINE ARRANGEMENT SEPARATED BY LESS THAN 2 FT'
  FORMAT(1,1) ENGINE, CR TWO ENGINES SEPARATED BY LESS THAN 2 FT'
+T6, 1 ONE ENGINE, CR TWO ENGINES SEPARATED CVER 2 FT'
+T6, 2 TWO ENGINES SEPARATED CVER 2 FT'
+T6, ENTER THE DESCRIPTION NUMBER IN I1 FCRMAT'
  READ(5,1211)I1
  IFC=I1 230
  GCNT INDE
  CALL FRTCMS(1,CLRSCRN *)
  WRIT(1,1) ENGINE PROTECTION '
  FORMAT(1,1) NONE PROTECTION AND/OR OVER 6 FT OF SEPARATION'
+T6, 1 WITH THE DESCRIPTION NUMBER IN I1 FCRMAT'
+T6, 2 ENTER WITH THE DESCRIPTION NUMBER IN I1 FCRMAT'
  READ(5,1211)I1
  IFC=I1 230
  GCNT INDE
  CALL FRTCMS(1,CLRSCRN *)
  WRIT(1,1) CONTROL SYSTEM POINT FAILURE (SPF) SITES'
  FORMAT(1,1) NO BACKUP - COVER 5 SINGLE SITES'
+T6, 1 NO BACKUP - UNDER 5 SPF SITES'
+T6, 2 WITH BACKUP - UNDER 5 SPF SITES'
+T6, 3 WITH BACKUP - OVER 5 SPF SITES'
+T6, 4 WITH BACKUP - UNDER 5 SPF SITES'
+T6, 5 NO SINGLE POINT FAILURE SITES'
+T6, ENTER THE DESCRIPTION NUMBER IN I1 FCRMAT'
  READ(5,1211)I1
  IFC=I1 230
  GCNT INDE
  CALL FRTCMS(1,CLRSCRN *)
  WRIT(1,1) CREW ARRANGEMENT
  FORMAT(1,1) NO BOTTOM SHIELD FOR PILOT BY ARMOR OR EQUIPMENT'
+T6, 1 NO BOTTOM SHIELD FOR PILOT BY ARMOR OR EQUIPMENT'
+T6, 2 NO SIDE SHIELD FOR PILOT BY ARMOR OR EQUIPMENT'
+T6, 3 NO PARTIAL ARMOR PROTECTION WITH STANCOFF(FRONT AND/CR BOTTO
  GM)'
+T6, 4 PARTIAL ARMOR PROTECTION WITH NO STANDOFF(FRONT AND/CR BO
  TTOM)'
+T6, 5 FULL ARMOR PROTECTION WITH STANDOFF (FRONT, BOTTOM, AND SI

```

```

&DES1, // FULL ARMOR PROTECTION WITH NO STANDOFF (FRONT, BOTTOM AND
+T6, //
&SIDES1, //
+T6, // ENTER THE DESCRIPTION NUMBER IN I1 FORMAT')
      YCA=I1
      GC TO 220
C*****
C MENU 31 *****
C ***** CALL FRTCMS('CLRS CRN ') *****
310 *****
31 *****
WRITE(4,1310)XMR,XMA,XMT //
FORMA *****
+T6, // 1 MISSION DESCR (31) *****
+T6, // 2 MISSION RADIUS OF ACTION, T51, F6.0, MI: //
+T6, // 3 MISSION LOITER ALTITUDE, T51, F6.0, FT: //
+T6, // 4 MISSION TIME ON STATION, T51, F6.0, MINUTES, //
+T6, // 5 CHANGE THEM ENTER ITS NUMBER IN I1 FORMAT. //
      IF(11.EQ.0) GO TC 120
      GO TO (311,312,313),I1
      WRITE(4,1310)
      GO TO 310
      WRITE(4,1311) MISSION RADIUS RANGE 200. TO 300. NM //
      FORMAT(7,F0, //
      FENTER THE NEW VALUE IN REAL NUMBER FORMAT. //
      FREAD(5,1202)I1
      XMR=V1
      GC TC 310
      WRITE(4,1312) LOITER ALTITUDE RANGE 1. TO 10000. FT //
      FORMAT(4,F0, //
      FENTER THE NEW VALUE IN REAL NUMBER FORMAT. //
      FREAD(5,1202)I1
      XMA=V1
      GC TC 310
      WRITE(4,1313) TIME ON STATION RANGE 60. TO 120. MIN. //
      FORMAT(4,F0, //
      FENTER THE NEW VALUE IN REAL NUMBER FORMAT. //
      FREAD(5,1202)I1
      XMT=V1
      GC TO 310
C***** MENU 32 *****
C ***** THREAT DEFINITION *****

```

```

+T6,, THE PROBABILITY OF HIT IS CEINED SEPARATELY FOR EACH.//
+T6,, AIRCRAFT AND THREAT. HOWEVER, THE FORM IS CONSISTENT WHERE.//
+T6,, P(H) REFERS TO THE PROBABILITY THAT A NON-MANEUVERING A/C.//
+T6,, WOULD BE HIT BY THE THREAT. F(M) IS THE MANEUVER FACTOR.//
+T6,, AND F(C) IS THE COUNTERMEASURE (CHAFF OR FLARE) FACTOR.//
+T6,, P(H) = P(H) * F(M) * F(C)
+T6,, ENTER ANY INTEGER TO RETURN TO MENU 4
READ(5,*)IJK
GO TO 130
9975 CALL FRTCMS('CLRSCRN ')
7975 WRITE(4,7975)
+T6,, THE VULNERABILITY ASSESSMENT SECTION CALCULATES //
+T6,, EITHER THE P(K/H) OR THE AVERAGE VULNERABLE AREA FOR THE //
+T6,, AIRCRAFT VERSUS A THREAT. FOR THE SUPPORT AIRCRAFT; //
+T6,, VS AAA //
+T6,, VS AV //
+T6,, = REGRESSION FORMULA THAT IS A FUNCTION //
+T6,, CF VULNERABILITY FEATURE INPUTS //
+T6,, P(K/H) = 1 - EXP (-1*AV/125.) //
+T6,, VS SAM //
+T6,, P(K/H) = FROM TABLE BASED UPON VULN. FEATURES //
+T6,, AV = 100. * EXP ( PKH - 1.) //
+T6,, ENTER ANY INTEGER TO RETURN TO MENU 5
READ(5,*)IJK
GO TO 140
9976 CALL FRTCMS('CLRSCRN ')
7976 WRITE(4,7976)
+T6,, THE FOLLOWING METHODS ARE USED FOR THE SUPPORT P(D).//
+T6,, VS A/A GUKS/MISSILE //
+T6,, VS P(D) = 1. //
+T6,, VS SAM //
+T6,, VS P(D) = TWO TIMES THE INTEGRAL OF THE GAUSSIAN //
+T6,, PROBABILITY FUNCTION FROM INFINITY TO CPA.//
+T6,, ENTER ANY INTEGER TO RETURN TO MENU 41
READ(5,*)IJK
GO TO 410
9977 CALL FRTCMS('CLRSCRN ')
7977 WRITE(4,7977)
+T6,, THE FOLLOWING METHODS ARE USED FOR THE SUPPORT P(H).//
+T6,, PH = PRCBABILITY THAT A NON-MANEUVERING A/C IS HIT //
+T6,, FA = PRCBABILITY THAT THE CREW IS ALERTED AND TAKES EVASIVE //
+T6,, ACTION. //
+T6,, FM = MANEUVER FACTOR //
+T6,, WS = WING LOADING/100. //
+T6,, VS AAA //
+T6,, VS AAA PF = .1456 //
+T6,, FC = CHAFF FACTOR //
+T6,, TW = THRUST TO WEIGH //
+T6,, FA = 1. //

```

```

7972 WRITE(4,7972)
FORMAT(1,1)
+T6, : DESIGN SECTION IS DIVIDED INTO THREE SUBSECTIONS.
+T6, : THE AIRCRAFT PERFORMANCE INDICATOR AND AVOID THE THREAT.
+T6, : ABILITY OF THE AIRCRAFT TO MANEUVER AND AVOID THE THREAT.
+T6, : SOME OF THE VALUES COMPLETELY. A FURTHER DISCUSSION OF THE
+T6, : THREAT OF ENVELOPE VALUES MAY BE FOUND IN THE P(H) SECTION.
+T6, : OF SUSCEPTIBILITY EVALUATION. FEATURES OF THE DESIGN ARE
+T6, : SUSCEPTIBILITY REDUCTION. FEATURES OF THE DESIGN ARE
+T6, : REDUCTION IN THIS SECTION. THESE INCLUDE JAMMER SIZE, RCS,
+T6, : INTERFERENCE LEVELS, CHAFF, DISPENSER AND RADAR WARNING RE-
+T6, : CEIVER. THE DEFAULT VALUES (BASELINE) ARE ZERO, INDICATING
+T6, : NCNE CF FEATURES ARE INCLUDED. FEATURES VARY WITH THE THREE
+T6, : TYPES OF AIRCRAFT DESIGN. THOSE FEATURES THAT BEST
+T6, : DESCRIBE YOUR DESIGN. MINIMUM VALUES OF 1 (BASELINE)
+T6, : INDICATE NO IMPROVEMENTS.
READ(5,*)IJK
GO TO 110

```

```

9973 CALL FRTCMS('CLRSCRN ')
7973 WRITE(4,7973)
FORMAT(1,1)
+T6, : THE COMBAT SCENARIO SECTION IS DIVIDED INTO TWO SUBSECTIONS.
+T6, : DEFINE IN THE SCENARIO VALUES APPLIED TO SPECIFICALLY
+T6, : BY THE SELECTION OF AIRCRAFT TYPE. THESE INCLUDE ITEMS
+T6, : THAT MIGHT BE CONSIDERED TACTICAL.
+T6, : IN THREAT SELECTION, THE THREAT DENSITIES AND THREAT
+T6, : DIAMETERS ARE ENTERED. THE THREATS FOR THE CLOSE AIR
-----
+T6, : SUPPORT MISSION ARE: AAA (OPTICLE), AAA (RADAR), AND
+T6, : LOW ALTITUDE SAM.
+T6, : ENTER ANY INTEGER TO RETURN TO MENU 3
READ(5,*)IJK
GO TO 120

```

```

9974 CALL FRTCMS('CLRSCRN ')
7974 WRITE(4,7974)
FORMAT(1,1)
+T6, : SUSCEPTIBILITY ASSESSMENT SECTION HAS TWO SUBSECTIONS.
+T6, : THE PROBABILITY OF DETECTION IS AFFECTED BY THE
+T6, : OF THE AIRCRAFT, THE POWER OF THE FINCISE JAMMER,
+T6, : AND THE SLANT RANGE FROM THE THREAT TO THE A/C AT CPA.
+T6, : NOTE THAT ALL AIRCRAFT ARE CONSIDERED TO PASS OVER A POINT
+T6, : THAT IS THE SAME HORIZONTAL DISTANCE FROM THE THREAT
+T6, : AS THE ALTITUDE OF THE AIRCRAFT. THIS MEANS THAT THE
+T6, : SLANT RANGE IS 1.414 TIMES THE ALTITUDE.

```

```

1656 +RMAT WRITE(4,1656)
FCRMT(0,16,ENTER MAX NUMBER OF SCRTIES FOR REPAIR IN I2 FC
READ(5,1697)I2
NS=I2
GC TO 659

699 CONTINUE
ARD=AAAD/FLCAT(JAM+1)
ADD=AAAD-ARC
CALL CAMP(XMR,AAAH,ARC,PKHAAA,FSAR,
&XMR,AAH,ADL,PKHAAA,PSAD,XMR,SAMH,SAMD,PKHSM,PSM,
&ACR1,NSRT,ICMS,CLRSCRN, TOTSR,TOTACK,TOTACL,TOTACR)
CALL WRITE(4,1659)ACR2,ICRSR,TOTACK,TOTACL,TOTACR
1699 FORMAT(1,UNDAMAGED AIRCRAFT,151,F6.0/
+T6, SCRTIES FLOWN,151,F6.0/
+T6, TARGETS ATTACKED,151,F8.0/
+T6, AIRCRAFT LOST,151,F8.0/
+T6, AIRCRAFT DAMAGED,151,F6.0/
+T6, TO RETURN A CAMPAIGN ENTER I,151,F6.0/
+T6, TO RETURN TO MENU (6) ENTER O,1)
READ(5,1211)I1 GO TO 65C
IF(I1.EQ.1) GO TO 150
GO TO 150
C*****
C HELP *****
C CALL FRTCMS('CLRSRN ') *****
9971 FORMAT(4,7971) *****
7971 +T6, THE VISAP DESIGN EVALUATOR IS DIVIDED INTO FIVE SECTIONS.//
+T6, THE AIRCRAFT IS ENTERED. THIS SECTION IS WHERE A DESCRIPTION OF//
+T6, THE AIRCRAFT IS ENTERED AS WELL AS DESCRIPTIONS OF THE S/V FEATURES.//
+T6, CONTAINED IN THE DESIGN VALUES. VALUES SHOWN INITIALLY ARE DEFAULT.//
+T6, VALUES WHICH MAY BE CHANGED.//
+T6, THE MISSION IS WHERE THE MISSION PARAMETERS AND//
+T6, THE INTENSITY VALUES ARE ENTERED. NOTE THAT THE TYPE OF THE//
+T6, OF THREAT DETERMINES THE THREATS. BECAUSE THE SELECTION OF THE//
+T6, THE LAST THREE SECTIONS ARE FOR EVALUATION OF THE//
+T6, DESIGN. IF THE DESIGN AND THREAT SECTIONS ARE NOT ENTERED//
+T6, DEFAULT VALUES (BASELINE) WILL BE USED FOR ALL CALCULATIONS.
+//T6,*)IJK
+READ(5,100)
GO TO 100
9972 CALL FRTCMS('CLRSRN ')

```



```

+ T6, TC RETURN TC MENU (6) ENTER 0' )
+ READ (5, 1211) I1
+ IF (11, EQ, 1) GO TO 640
GO TO 150
C*****
C MENU 63 CAMPAIGN ANALYSIS
C*****
650 CONTINUE
659 CALL FRTCMS (CLRSCRN, )
1650 WRITE (4, 16) XNPASS, NS, PSAR, PSAD, PSSM
      WRITE (4, 17) CAMPAIGN ANALYSIS, I46, F6.0/
      WRITE (4, 18) MEFT IN CAMPAIGN, I43, I6/
      WRITE (4, 19) AIR CRAR IN RAIDS IN PER SORTIE, I43, F6.0/
      WRITE (4, 20) NUMBER OF PASSES FOR REPAIR, I43, I6.4/
      WRITE (4, 21) NUMBER OF SORTIES, I48, F6.4/
      WRITE (4, 22) P(S) VS AAA(OPTICAL), I48, F6.4/
      WRITE (4, 23) P(S) VS AAA(CPTICAL), I48, F6.4/
      WRITE (4, 24) P(S) VS LOW ALT SAM, I48, F6.4/
      WRITE (4, 25) P(S) VS CHANGE A VALUE ENTER ITS NUMBER, I48, F6.4/
      WRITE (4, 26) P(S) VS CHANGE Q FOR NO CHANGE ENTER REQUIRED, I48, F6.4/
      READ (5, 11) I1
      IF (11, EQ, 0) GO TO 699
      IF (11, NE, 1) GO TO 651, 652, 653, 656, 618, 628, 638, I1
      GO TO 1200
1658 WRITE (4, 1) TO 650
      CONTINUE
651 WRITE (4, 0, 16, 1651) ENTER NUMBER OF A/C IN REAL NUMBER FORMAT.
1651 FORMAT (10, 16, 1651)
1657 FORMAT (5, 1657) V1
      FORMAT (18, 4)
      ACRI = V1
      CONTINUE
652 WRITE (4, 0, 16, 1652) ENTER NUMBER OF RAIDS IN I2 FORMAT.
1652 FORMAT (10, 16, 1652)
1697 READ (5, 1697) I2
      FORMAT (12)
      NS TO 655
      CONTINUE
653 WRITE (4, 1, 16, 1653) ENTER PASSES PER SORTIE IN REAL NUMBER FORMAT.
1653 FORMAT (10, 16, 1653)
      FOR READ (5, 1202) V1
      XNPASS = V1
      GO TO 659
      CONTINUE
656

```



```

611      GC TO 618
        CCNTINUE
        WRITE(4,1611)
1611     FORMAT('0',T6,'ENTER P(D),P(H),P(K/H) IN REAL NUMBER FORMAT.'/
        +T6,
        READ(5,1657)PDAR,P(D),P(H),P(K/H) IN REAL NUMBER FORMAT./
        PSAR=1.-POAK*PHR*PKHAA
        GO TO 618
C*****
C(S) AAA OPTICAL
CALL FR(CMS,CLRSCRN,1)
620     PSAR = 1.-PDAR*PHR*PKHAA
628     WRITE(4,2220)JAM,IFS,IRCS,IFV,IMARN,IFF,ICHAFF,IEA,IEP,ICS,ICA
        FORMAT('1',T6,T6, ' THE PROB OF SURVIVAL VS AAA(OPTICAL)')//
1620     WRITE(PS,1620)FSAO,PCAO,PHO,PKHAAA
        +T6,PS = 1.-PDAR*PHR*PKHAA
        +T9,F6.4,7X,F6.4,2X,F6.4,3X,F6.4//
        +T6, ' CHANGE THIS VALUE ENTER 1 IN 11 FORMAT.'/
        +T6, ' ENTER 0 FOR NO CHANGE REQUIRED.'/
        READ(5,1659)II
        IF(II.EQ.0) GO TO 150
        GO TO 1621
1629     WRITE(4,1629),II
        CCNTINUE
621     WRITE(4,1611)
        READ(5,1657)PDAR,P(D),P(H),P(K/H) IN REAL NUMBER FORMAT./
        PSAR=1.-POAK*PHR*PKHAA
        GO TO 628
C*****
C(S) LOW ALI SAM
PSSM = 1.-PCSM*PHSM*PKHSM
630     CALL FR(CMS,CLRSCRN,1)
638     WRITE(4,2220)JAM,IFS,IRCS,IFV,IMARN,IFE,ICHAFF,IEA,IEP,ICS,ICA
        FORMAT('0',T6, ' THE PROB OF SURVIVAL VS SAM')//
1630     WRITE(PS,1630)FSAO,PCAO,PHO,PKHAAA
        +T6,PS = 1.-PDAR*PHR*PKHAA
        +T9,F6.4,7X,F6.4,2X,F6.4,3X,F6.4//
        +T6, ' CHANGE THIS VALUE ENTER 1 IN 11 FORMAT.'/
        +T6, ' ENTER 0 FOR NO CHANGE REQUIRED.'/
        READ(5,1659)II
        IF(II.EQ.0) GO TO 150
        GO TO 1631
1639     WRITE(4,1639),II
        CCNTINUE
631

```

```

1521 CALL FRTCMS('CLRSCRN')
1522 WRITE(4,1522) VASM, COMPUTED A(V) VS SAM IS 'T51,F6.0,' SQFT' /
      FORMAT(10,16,' THE P(K/H) IS 'T51,F6.4' /
+T6,' TC CHANGE FCR NE THESE VALUES ENTER 1' /
+T6,' ENTER FCR NE CHANGE REQUIRED')
      READ(5,121) I1
      IF(I1.EQ.0) GO TO 14C
      IF(I1.EQ.1) GO TO 1523
      WRITE(4,120C)
      GO TO 1521
1524 GO CONTINUE(1525)
1523 WFORMAT(4,10,' VULN AREA RANGE 37.0 TO 100.C '//
+T6,' ENTER THE NEW VALUE IN REAL NUMBER FORMAT.')
      READ(5,1202) I1
      VASM=V1
      PKHSM = 1. + ALOG(VASM/100.)
      GC TO 1521
C***** P(S) AAA RADAR *****
C MSNU 61 *****
C ***** PSAR = 1 - PDAR * PHR * PKHAAA *****
610 CALL FRTCMS('CLRSCRN')
618 WRITE(4,2220) JAM, IFS, IPCS, IFV, IWARN, IFE, ICHAFF, IEA, IEP, ICS, ICA
      FORMAT(10,16,' SUSCEPTIBILITY REDUCTION FEATURES '//
+T40,' JAMMER ABILITY REDUCTION FEATURES '//
+T42,' JAMMER NUMBER GENERAL 'T51,11' /
+T42,' RCS REDUCTION LEVEL 'T35,11' /
+T42,' FUEL/VOLIC INTERFACER 'T35,11' /
+T42,' RADAR/WARNING INTERFACE 'T35,11' /
+T42,' CHAFF DISPENSEMENT 'T35,11' /
+T42,' ENGINE PROTECTION 'T69,11' /
+T42,' CONTROL SYSTEM 'T69,11' /
+T42,' CREW ARRANGEMENT 'T69,11' /
      WRITE(4,1610) PSAR, FCFAR, PHR, PKHAAA
      FORMAT(10,16,' THE PROB OF SURVIVAL VS AAA(RADAR) '//
+T6,' F6.4, 7X, F6.4, 3X, F6.4, 3X, F6.4' /
+T9,' TC CHANGE THIS VALUE ENTER 1 IN I1 FORMAT' /
+T6,' ENTER 0 FCR NO CHANGE REQUIRED')
      READ(5,1659) I1
      IF(I1.EQ.0) GO TO 150
      GO TO (611,1619), I1
      WRITE(4,120C)
1610 WRITE(4,120C)

```

```

423      CONTINUE
1423      CALL SRPHSM(IWARN, ICHAFF, TW, WS, PHSM)
         CALL FRTCMS(:CLRSCRN, :)
14923     WRITE(4,14923)PHSM
         FORMAT(1,'T6',THE PRCB. OF HIT BY LOW ALT. SAM IS '
         +T6,'ENTER O FCR NC CHANGE REQUIRED')
         READ(5,1211)I1
         IF(I1.EQ.0) GO TO 420
         IF(I1.EQ.1) GO TO 14232
14933     WRITE(4,1423)
14232     GO CCNTINUE
         WRITE(4,1497)
         READ(5,1202)V1
         PHSM=V1
         GO TO 1423
C*****
C      MENU 51  VULN. AREA / P(K/H) VS AAA *****
C*****
510      CONTINUE
         CALL SRVAAA(IFS, IFV, IFE, IEA, IEP, ICS, ICA, TW, WS, XMR, XMA, XMT, WT, VAA
         &AA, PKFAAA)
1511     CALL FRTCMS(:CLRSCRN, :)
         WRITE(4,1512)VAAAA,PKHAAA
1512     FORMAT(1,'T6',THE CCMPUTED A(V) VS AAA IS ',T51,F6.0,' SQRT',/
         +T6,'THE P(K/H) IS ',T51,F6.4,/)
         +T6,'TO CHANGE THESE VALUES ENTER 1'
         +T6,'ENTER C FCR NO CHANGE REQUIRED')
         READ(5,1211)I1
         IF(I1.EQ.0) GO TO 140
         IF(I1.EQ.1) GO TO 1513
1514     WRITE(4,1511)
1513     GO CCNTINUE
1515     +T6,'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.'//
         WRITE(4,1515)
         FCRMAT(0,'
         +T6,'ENTER THE NEW VALUE IN REAL NUMBER FORMAT.'//
         VAAAA=V1
C- #1 -----
         PKHAAA = 1. - EXP(-1.*VAAAA/125.)
         GO TC 1511
C*****
C      VULN. AREA / P(K/H) VS LOW ALT SAM *****
C*****
520     CCNTINUE
         CALL SRVASM(IFS, IFV, VASM, PKHSM)

```

```

C*****
C      IF(K8Q.EQ.KK(4)) GC TC 130
C      WRITE(4,1200)
C      GO TO 42
C*****
C      PROB CF HIT AAA RADAR
C*****
C      421 CONTINUE
C          CALL SRPHC(TW,MS,PHR)
C          CALL FRTCMS(,CLRSCRN,')
C          WRITE(4,14911)PHR
C      14911 FORMAT(1,16,THE COMPUTED P(H) BY AAA(RADAR) IS
C          +16,ENTER 0 FOR NC CHANGE THIS VALUE ENTER 1,
C          +16,READ(5,1211)I1
C          IF(I1.EQ.0) GC TC 42C
C          IF(I1.EQ.1) GO TO 14212
C      14913 WRITE(4,12CC)
C      14212 GO CONTINUE
C          WRITE(4,1497)
C          FORMAT(4,C,PH RANGE 0.0 TO 1.0 '//
C      1497 +16,ENTER THE NEW VALUE IN REAL NUMBER FORMAT.)
C          READ(5,1202)VI
C          PHR=VI
C          GO TO 1421
C*****
C      PROB CF HIT AAA OPTICAL
C*****
C      422 CONTINUE
C          CALL SRPHC(TW,PHO)
C          CALL FRTCMS(,CLRSCRN,')
C          WRITE(4,14921)PHC
C      14921 FORMAT(1,16,THE COMPUTED P(H) BY THE AAA(OPTICAL) IS
C          +16,ENTER 0 FOR NC CHANGE THIS VALUE ENTER 1,
C          +16,READ(5,1211)I1
C          IF(I1.EQ.0) GO TO 42C
C          IF(I1.EQ.1) GO TO 14222
C      14929 WRITE(4,12CC)
C      14222 GO CONTINUE
C          WRITE(4,1497)
C          READ(5,1202)VI
C          PHC=VI
C          GO TO 1422
C*****
C      PROB CF HIT LGW ALT SAM
C*****
C*****

```

```

IF(I1.EQ.0) GO TC 410
IF(I1.EQ.1) GO TC 1416
1418 WRITE TO 4120C)
GO
1416 CONTINUE
WRITE(4,1417)
READ(5,1202)V1
PCAO=V1
GC TO 412
*****
C PD VS SAM *****
C *****
413 CONTINUE
CALL SRPCMS ('CLRSCRN ')
1490 WRITE(4,1491)PDSM
1491 FORMAT('I',T6,'THE COMPUTED P(C) BY LOW ALT. SAM (RADAR) IS '
+T6,'ENTER C FOR NO CHANGE REQUIRED')
+T6,'EQ.0) GO TC 410
READ(5,1201)V1
IF(I1.EQ.1) GO TC 1492
1493 WRITE TO 4120C)
GO
1492 CONTINUE
WRITE(4,1417)
READ(5,1202)V1
PCSM=V1
GC TO 1490
*****
C MENU 42 PRCP OF HIT *****
C *****
420 CALL FRTCMS ('CLRSCRN ')
42 CONTINUE
1420 WRITE(4,1420)
FORMAT('I',MENU(42) SELECT A CODE AS FOLLOWS: '//
+T6,'FOR AN EXPLANATION)
+T6,'P(H) VS AAA (RADAR)
+T6,'P(H) VS LOW ALT. SAM
+T6,'TC RETURN TO MENU (4)
+T6,'TO TRANSFER TO OTHER MENUS
READ(5,2000)K8C
IF(K8C.EQ.K6(1)) GC TC 421
IF(K8C.EQ.K6(2)) GC TC 422
IF(K8C.EQ.K6(3)) GC TC 9977
IF(K8C.EQ.KK(1)) GC TC 998
IF(K8C.EQ.KK(2)) GC TC 998

```







```

+T6, . FM= 1 - .05 * TW
+T6, . PH = PH * (1. -(1.--FM)*FA)
+T6, . VS AAA (RADAR)
+T6, . PF= 1.
+T6, . FM= (1.--.15 * TW)*(1. - EXP(-I*(WS-50.)/10.))
+T6, . PH = PH * (1. -(1.--FM)*FA)
+T6, . VS SAM
+T6, . PF = .17512 FC = .6
+T6, . FM = 1.-7.80516*TW + 6.166 * WS**8 + .907024 * WS**-4
+T6, . PH = PH * FC * (1. -(1.--FM) * FA)
+T6, . ENTER ANY INTEGER TO RETURN TO MENU 42
+T6, . READ(5,*)IJK
+T6, . GO TO 420
9978 GO CALL FRTCMS('CLRSCRN ')
9978 WRITE(4,7978)
+T6, . THE P(S) VS INDIVIDUAL WEAPONS HELP FCR MENU 6
+T6, . THE P(S) = 1 - P(D) * P(H) * P(K/H)
+T6, . THE P(S) FOR SINGLE SORTIE
+T6, . WH = WEIGHTING FACTOR
+T6, . XH = A/C HIT XK = A/C KILLED A = A/C DAMAGED
+T6, . H = A/C OVER TARGET AC = A/C DAMAGED
+T6, . ACOVER = PRCB. CF MISS. SURVIVAL PSM = PS ** W
+T6, . WH = XL * XH * C / 100.
+T6, . PH = (1. - PSM)/PKH
+T6, . XH = H * PKH
+T6, . ACOVER = ACRI - H1 - H2 - H3
+T6, . H4 = ACRI - H1 - H4
+T6, . A3 = H4 - XK4
+T6, . ACDA = A1 + A4
+T6, . ENTER ANY INTEGER TO RETURN TO MENU 6
+T6, . READ(5,*)IJK

```

```

C- #3 GO TO 150
C- #5 ***** PLOTTING INFORMATION *****
8988 CALL FRTCMS('CLRSCRN ')
8889 WRITE(4,8889)
+T6, . MFNU (8) INCORPORATES A DATA GENERATING ROUTINE TO SAVE
+T6, . THE PROBABILITIES OF A KILL (PK) = 1 - P(S) FOR
+T6, . LATER PLCTTING. THIS PLOT MAY BE OBTAINED FROM A
+T6, . TEKTRONIX DUAL SCREEN APPROPRIATE THREAT TYPES ARE PRESENTED.
+T6, . AGAINST THE APPROPRIATE THREAT TYPES ARE PRESENTED.
+T6, . YOUR BASELINE DESIGN. YOUR FIRST DESIGN, AND ONE OTHER
+T6, . MODIFICATION. YOU MUST HAVE DCNE AND CHCSEN TO SAVE

```









```

XX(JJ+1)=XI
PD=PD+AREA
10 CONTINUE
   IF(PCF.GT.0.01) GO TO 20
   PDF=.100
   CONTINUE
20  RETURN
END
SUBROUTINE SRPHO(TWS,PDAM)
C*****
C***** AAA OPTICAL PH *****
C*****
PH = 1.456
FM = (1.-.05 * TWS)
XMF = 1.-FM
PDAM = PH*XMF
RETURN
END
SUBROUTINE SRPHR(TWS,WSS,PCAM)
C*****
C***** AAA RADAR PH *****
C*****
PH = 1.
FM = (1.-FM)*FA
XMF = 1.-FM
PCAM = PH*XMF
RETURN
END
SUBROUTINE SRPHSM(IWARN,ICHAFF,TWS,WSS,PDAM)
C*****
C***** XXMR = 1.5 *****
C***** SAM PH *****
C*****
PH = .15
CALL SRFC(IWARN,ICHAFF,FC)
C*****
C***** MODIFIED FOR CHAFF *****
C*****
PH1 = PH * FC
CALL SRFA(XXMR,IWARN,FA)
C*****
C***** MODIFIED FOR MANUVERING *****
C*****
FM = 1.-7.80516*TWS+6.166*(WSS/100.)*.8+.907024*(100./WSS)**4

```

```

IF(FM.LT.0.01)FM = 0.01
PCAM = PF1 * XMF
RETURN
END
C***** SURROUTINE SRFA(XXMRS,IMARN,FAS) *****
C***** ALERTICN FACTOR *****
REAL DATA MRM(28),FVM(28),FESM(12),MRSM(12)
DATA MRM/0.,.36,.40,.50,.60,.70,.80,.90,.100.,110.,120.,140./
DATA FESM/0.,.254,.2.,.2.54,.3.,.3.08,.3.48,.3.78,.4.34,.4.23
DATA MRM/0.,.1.55,.4.68,.4.8.,.4.9.,.5.05,.5.35,.5.8.,.6.,.6.39,.7.,.7.36
A,.4.,.35,.4.,.45,/.
B DATA FVM/1.,.995,.55.,.543,.9.,.868,.85.,.8.,.75.,.712,.7.,.65.,.6
A IF(IMARN.EQ.1)GO TO 15
KEY=1
CONTINUE
IF(MRM(KEY).LT.XXMRS) GO TO 5
DELTAX=MRM(KEY)-FVM(KEY-1)
FVS=(XXMRS-MRM(KEY-1))/DELTAX)*DELTAX+FVM(KEY-1)
FAS = FVS
GO TO 20
CONTINUE
J=1
CONTINUE
IF(J+1)
IF(XXMRS.GE.MRSM(J)) GO TO 10
DX = MRSM(J-1)-MRSM(J)
FES=(XXMRS-MRSM(J-1))/DX)*DX+FESM(J-1)
FAS = FES
CONTINUE
END
SUBROUTINE SRFC(ICPAFF,FC)
REAL PBTSM(17)
DATA PBTSM/.00,.19,.35,.49,.6.,.68,.74,.8.,.83,.86,.9.,.92,.935,
.95,.96,.97,.98/
A PBT5=0.
IF(ICPAFF.EQ.0) GC TO 10
NBUNDS = 4
PBT5=PBTSM(NBUNDS+1)
FC = 1. - PBT5
RETURN
10
15
20

```



```

ENC
SUBROUTINE SRVASM(IFSS,IFVS,VASMS,PKHSMS)
*****
***** VULNERABLE AREA VS SAM *****
***** SET UP VALUES *****
***** DATA F1/GE/3, F2/0.7, F3/0.7, F4/0.7 *****
***** IF((IFSS.NE.1).AND.(IFSS.NE.3)).AND.(IFSS.NE.6)) F2 = 1. *****
***** IF((IFVS.EQ.3)).OR.(IFVS.EQ.4)).CR.(IFVS.EQ.6)) F3 = 1. *****
***** IF((IFVS.EQ.5)).OR.(IFVS.EQ.6)) F4 = 1. *****
***** CALCULATE PK/H VS SAM *****
***** ITF = INT(F1+F2+F3+F4) *****
***** GO TC (10,20,30,40),ITF *****
***** GO PKHSMS = 1. *****
***** GO TC 50 *****
***** GO PKHSMS = F1*.584 + F2*.995 + F3*.668 + F4*.911 *****
***** GO TC 50 *****
***** GO PKHSMS = F1*F2*.984 + F1*F3*.521 + F1*F4*.898 + *****
***** GO PKHSMS = F2*F3*.628 + F2*F4*.904 + F3*F4*.324 *****
***** GO TC 50 *****
***** GO PKHSMS = F1*F2*F3*.484 + F1*F2*F4*.896 + *****
***** GO PKHSMS = F1*F3*F4*.223 + F2*F3*F4*.258 *****
***** GO TC 50 *****
***** GO PKHSMS = .182 *****
***** CONTINUE *****
***** CALCULATE VULN AREA VS SAM *****
***** VASMS = 100. * EXP(PKHSMS-1.) *****
RETURN
END
SUBROUTINE SRVAAA(IFSS,IFVS,IFES,IEAS,IEPS,ICSS,ICAS,A,B,C,D,E,F,
*****
***** EVAAAAS,PKFAAS) *****
***** VULNERABLE AREA VS AAA *****
***** DIMENSION XFE(8),XEA(2) *****
***** DATA XFE/2.73,4.6,8.12,16.,32./ *****
***** DATA XEA/2.7,1./ *****
***** WRS = 11333. + 1.34555*AA*F - .00247249*RR*F + .067473*CFE *****
***** WRS = -11.1186*AB + 18.6825*AC + 3591*AAF + .236943*BE *****
***** GO TC 50 *****
***** GO PKHSMS = - * COORC00441459*D*F + .00256493*EF *****
***** FT = 6 * SET UP VALUES *****
***** FS = FLCAT(IFSS) *****
***** FF = XFE(IFES) * FT * .001 *****
***** EA = XEA(IEAS) * WES * .001 *****

```

```

EP = FLCAT(IIEPSI)
CS = FLCAT(ICASSI)
CA = CALCULATE
VAAAAS = 41.56 - 2.244*ALOG(FE) - 4.373*ALOG(FV) - 4.732*ALOG(FS)
          + 5.009*ALOG(CS) + 5.946*ALOG(EA) - 2.491*ALOG(CA)
          + 16.44*ALOG(FT*.001) - 47.503*ALOG(EPI)
***** CALCULATE P(K/H) *****
C- #1 ----- PKPAAS = 1.- EXP (-1.*VAAAAS/125.)
RETURN
END
SUBROUTINE SORTI
  XL1,XH1,D1,PKH1,PS1, XL2,XH2, D2,PKH2,PS2, XL3,XH3,D3,PKH3,PS3,
  &ACR1,NSRT,XNPAS,NS, &ACR2,TCTSR,TOTACK,TOTACL,TCTACR)
*****
XL-PENDIS F-THREAT DENSITY D-THREAT DIAMETER
*****
W1 = XL1 * XH1 * D1 / 100.
W2 = XL2 * XH2 * D2 / 100.
W3 = XL3 * XH3 * D3 / 100.
PSM1 = PS1 ** W1
PSM2 = PS2 ** W2
PSM3 = PS3 ** W3
PH1 = (1. - PSM1) / PKH1
PH2 = (1. - PSM2) / PKH2
PH3 = (1. - PSM3) / PKH3
*****THREAT 1*****
H1 = ACR1 * PH1
XK1 = H1 * PKH1
A1 = F1 - XK1
*****THREAT 2*****
H2 = (ACR1-F1) * PH2
XK2 = F2 * PKH2
A2 = H2 - XK2
*****THREAT 3*****
H3 = (ACR1-H1-H2) * PH3
XK3 = H3 * PKH3
A3 = H3 - XK3
*****COVER *****
ACOVER = ACR1-F1-H2-H3
ATAC = ACOVER * XNPAS
*****EGRESS*****

```

```

C*****THREAT 1*****
H4 = ACQVER PKH1
XK4 = H4 - XK4
A4 *****
C*****THREAT 2*****
H5 = (ACQVER-H4) * PH2
XK5 = H5 * PKH2
A5 = H5 - XK5
C*****THREAT 3*****
C- #1 -----
H6 = (ACQVER-H4-H5) * PH3
XK6 = H6 * PKH3
A6 *****
C*****TOTALS FOR SORTIE*****
ACNHT = ACRI-H1-H2-H3-H4-H5-H6
ACDAM = A1 + A2 + A3 + A4 + A5 + A6
ACKIL = XK1 + XK2 + XK3 + XK4 + XK5 + XK6
C*****FDR NEXT SORTIE*****
ACR2 = ACNHT
TOTSR = ACRI
TCTACK = ATAC
TOTACL = ACKIL
TCTACR = ACDAM

```

RETURN  
END

```

SUBROUTINE CAMPI *****
SUBROUTINE FOR CAMPAIGN ANALYSIS *****
*****
XLI, XH1, D1, PKH1, PS1, XL2, XH2, D2, PKH2, PS2, XL3, XH3, D3, PKH3, PS3,
&ACR1, DIMENSION ACR(200)
IF(NS.EC.GINS = 1
TOTSR = 0.
TCTACK = 0.
TCTACR = 0.

```

C W1 = XLI \* XH1 \* D1 / 100.  
W2 = XL2 \* XH2 \* D2 / 100.  
W3 = XL3 \* XH3 \* D3 / 100.

C PSM1 = PS1 \*\* W1  
PSM2 = PS2 \*\* W2  
PSM3 = PS3 \*\* W3

C PH1 = (1. - PSM1)/PKH1  
PH2 = (1. - PSM2)/PKH2

PH2 = (1. - PSM3)/PKH3

C ACR(I) = ACRI  
DO I = I INSR  
C \*\*\*\*\* INGRESS \*\*\*\*\*

H1 = ACRI \* PHI  
XK1 = H1 \* PKH1  
A1 = H1 \* XK1  
H2 = (ACR(I)-H1) \* PH2  
XK2 = H2 \* PKH2  
A2 = H2 \* XK2  
H3 = (ACR(I)-H1-H2) \* PH3  
XK3 = H3 \* PKH3  
A3 = H3 \* XK3

C \*\*\*\*\* COVER TARGET \*\*\*\*\*  
ACOVER = ACR(I)-H1-H2-H3  
ATAC = ACOVER \* XNPAS

C \*\*\*\*\* EGRESS \*\*\*\*\*  
H4 = ACCOVER \* PHI  
XK4 = H4 \* PKH1  
A4 = H4 \* XK4  
H5 = (ACCOVER-H4) \* PH2  
XK5 = H5 \* PKH2  
A5 = H5 \* XK5

C- #1 -----  
H6 = (ACOVER-H4-H5) \* PH3  
XK6 = H6 \* PKH3  
A6 = H6 \* XK6

C \*\*\*\*\* TICALS FCR SORTIE \*\*\*\*\*  
ACNHT = A1 + A2 + A3 + A4 + A5 + A6  
ACDAM = A1 + A2 + A3 + A4 + A5 + A6  
ACKIL = XK1 + XK2 + XK3 + XK4 + XK5 + XK6  
C \*\*\*\*\* FOR NEXT SORTIE \*\*\*\*\*

TCTACR = TOTACR+ACDAM  
TCTACR = TOTACR/FLOAT(NS)  
TCTACR = TOTACR-ACROUT  
ACR(I+1) = ACNHT + ACRGUT  
ACR2 = ACR(I+1)  
TOTSR = TOTSR + ACR(I)  
TOTACK = TOTACK + ATAC  
TOTACL = TOTACL + ACKIL

10 CCNTINUE  
RETURN  
END

C \*\*\*\*\* SUBROUTINE SSRWT (SPAN, W, DL, EC, EL, IM, MS, WT, XL, ICA, ICS, IEA, \*  
C \*\*\*\*\* SUPPORT A/C WEIGHT ESTIMATIONS ROUTINE \*\*\*\*\*  
C \*\*\*\*\*

```

@
IFP,IFE,IFS,IFV,JAM,IRCS,XMR,XVA,XMT,IWARN,ICHAFF,
BLTOGW,TCGW)
REAL A,B,C,D,E,F,I,J
A==XMR
B==XMA
C==XMT
D==WT
E==O.
F==O.
I==O.
J==O.
C*****
BLTOGW = .129616E+05 + .425125*F + 2.16928*H + 2.227*A*F
a+.16377*C*I + 0.27286*E - 0.2*E*F + 21.1066*F*G
a-.12935E-04*E*H + 1.672*F*J
C*****
A/C TCGW OF DESIGN WITH SURVIVABILITY ENHANCEMENT
*****
THE FOLLOWING ASSUMPTION MADE: 23 MM TANKS 1 PLUS SUMP TANK(S)
ALL SELF-SEALS DONE FUSELAGE TANKS 1
TWO SUMP TANKS 1 HAVE EQUAL VOLUME
EACH TANK HOLDS 1/7 OF TOTAL VOLUME
INTERNAL FOAM USE VICE INTERNAL FOAM
FIRE EXTINGUISHING VICE EXTERNAL FOAM
*****
IF (JA.EQ.C1 GO TO 40
IF (G = .8675 * W5 / BLTOGW
CONTINUE
*****
WEIGHT INCREASE CALCULATIONS
*****
FR = FUEL REQUIRED FOR MISSION *****
FR = -11.1182*AB + 186.6825*AC + 0.3591*AF + 6853.06*AJ
a+.0256492*AE + 186.885*CG - 21.0454*CI - 0.44145E-05*DF
a-.0.397397E-05*AH + 0.415838*H*J
*****
XNT = 2
XNT (IFS.EQ.1) OR (IFS.EQ.3) OR (IFS.EQ.6) XNT = 0.
IF (IFS.EQ.2) XNT = 1
WSSP = 1.45*12.2*8.77 - 1.1*(1.7) **.75*(FR/6.6) **.64*XNT **.11
*****
WF = 0
IF (IFV.EQ.4) OR (IFV.EQ.6) WF = .0186 * FR/6.6
*****

```

```

C***** WEIGHT INCREASE DUE TO FIRE EXTINGUISHING *****
WFE = 0.
XV = 4./3. * ( EC + 1.1 * EC * FL
IF((IFV.EQ.5).CR.(IFV.EQ.6)) WFE = 10.5 * XV*.26
C***** WEIGHT INCREASE DUE TO DUCT PROTECTION *****
XND = 1.
WNB = 0.
IF((IEP.EQ.2) XND = 2.
IF((IEP.EQ.2) XND * .5
XS = ECL * EC * XND * .5
IF((IFE.EQ.2).OR.(IFE.EQ.4).OR.(IFE.EQ.6)) WNB = 7.6 * XS
C***** AD *****
AD = 0.
IF((ICA.EQ.2) AD = 10.
IF((ICA.EQ.3).OR.(ICA.EQ.4)) AD = 18.
IF((ICA.EQ.5).CR.(ICA.EQ.6)) AD = 30.
WARM = 12. * AD
C***** WEIGHT INCREASE DUE TO ENGINE SEPERATION *****
XEB = 0.
IF((IEA.EQ.1) XEB = 0.
IF((IEA.EQ.2) XEB = 4.
IF((IEP.EQ.2) XEB = 6.
XNA = ED * 12.
XN = 11.
WENG = 12000.
WENG * XN * .526 * XA * XEB * .000001
C***** WES *****
WES = WENG * XN * .526 * XA * XEB * .000001
C***** WEIGHT *****
XS = 0.
IF((IRCS.EQ.1) XS = 10.
IF((IRCS.EQ.2) XS = 16.
IF((IRCS.EQ.3) XS = 16. + BLTDGW/WS *.69
WARM = XT * XS * 23.8
C***** WEIGHT INCREASE DUE TO REDUNDANT CONTROLS *****
BACKUP = 0.
IF((ICS.EQ.3).OR.(ICS.EQ.4)) BACKUP = 1.
XLGP = EC + SPAN + XL / 2.
WRED = BACKUP * (2.207 * XLGP - 4.79)
C***** WEIGH *****
WEIGH = 0.
IF((IWRM.EQ.1) WEW = 50.
IF((IWRM.EQ.2) WEW = 100.
IF((IWRM.EQ.3) WEW = 200.
IF((IWRM.EQ.4) WEW = 500.
IF((IWRM.EQ.5) WEW = 1000.
C***** WEIGHT INCREASE DUE TO CHAFF DISPENSER *****

```

```

C*****
WCD = 0.
IF (ICHAFF.EQ.1) WCE = 86.
WEIGHT INCREASE DUE TO SUBMERGED STORE *****
WSCR = 0.
IF (IRCS.EC.3) WSOR = 1.13 * WT/100.
*****
C***** TOTAL WEIGHT INCREASE *****
C***** H = WSSP + WFE + WBB + WARM + WES + WRAM + WRED + WFM + WJW + WCO + WSOR *****
C***** TOTAL TOGW ON ENHANCED A/C *****
C***** TOGW = 1.29616E+05 + .425125*F + 2.16928*H + 2.227*A*F *****
          @+.163377*C*E - 13.6801*C*I + 0.272868E-02*E*F + 21.1006*F*G *****
          @-.12925E-04*E*H + 1.672*F*J *****
RETURN
END

```

```

*****
C** SUPPLT < SUPPRT PLCT >
C** PROGRAM TC GRAPH P(K),S VS. THREATS AGAINST
C** CLOSE AIR SUPPRT AIRCRAFT
C**
*****
C** DIMENSION Y0(3),Y1(3),Y2(3),Y3(3),XC(3),IPKRAY(400)
C**
C** DATA LABEL/,AAA(,RA(,AAA(,OP,,TICA,,L(,
C**          ,SAM(,RA(,AAA(,OP,,TICA,,L(,
C** DATA YC/0.,0.,0./
C** DATA XC/1.,2.,3./
C
C 900 READ(5,900)(Y1(I),I=1,3),(Y2(I),I=1,3),(Y3(I),I=1,3)
C    FORMAT(F4.2)
C
CALL TFK618
CALL HWRCT('MOVIE')
CALL BLOWUP(1.4)
CALL PAGE(8.5,11.)
CALL AREA2D(6.,8.5)
CALL XNAME('THREATS',100)
CALL YNAME('F(K)',100)
CALL SWISSM(50.,1.,003,1)
CALL SHDCHR('STAND')
CALL BASALF('L/CSTC')
CALL MIXALF('L/CSTC')
CALL HEADIN('CLOSE AIR SUPPORT AIRCRAFT$,100,1.5,2)
CALL HEADIN('LOSS) R(ATE) VS. T(HREAT) T(TYPE)$,100,1.1,2)
CALL YAYANG(0.)
CALL XTICKS(0)
CALL XAXEND('NCEXDS')
CALL YAXEND('NOLAST')
CALL SWISSL
CALL HEIGHT(.12)
CALL FRAME(10)
CALL THKFRM(.10)
CALL XLABGR(LABEL,4,5,C.,1,1.1)
CALL BLREC(0.01,8.1,4,29,0.6,0)
CALL BLKEY(4,3,7.7,1.7,1.0,0)
CALL CLUSTR(3,0E)
CALL HEIGHT(.08)
CALL BARDC('SECCND', 'OUTSIDE',2)

```



```

CALL BLBAR (, LABEL, , YO, Y1, 3)
CALL VBAR (, LABEL, , YO, Y2, 3)
CALL VBAR (, LABEL, , YO, Y3, 3)
CALL HEIGHT (.05)
CALL CCT D(0, 2)
CALL RESET (, DOT)
CALL HEIGHT (.10)
CALL BLOFF (, C)
CALL IN=LINE ST (, IPKRAY, 400, 40)
CALL LINES (, B(ASELINE), , IPKRAY, 1)
CALL LINES (, 1ST L(ESIGN), , IPKRAY, 2)
CALL LINES (, 2ND L(ESIGN), , IPKRAY, 3)
CALL LEGEND (, IPKRAY, 3, 4.5, 7.6)
CALL ENDPL(0)
CALL CCNEPL
CALL STOP
END

```

## LIST OF REFERENCES

Hesser, N. P., The Development of an Interactive Computer Program for the Survivability Evaluation of Aircraft Conceptual Designs, M.S. Thesis, Naval Postgraduate School, Monterey, California, June 1982.

Ball, R. E., The Fundamentals of Aircraft Survivability Analysis and Design, Department of Aeronautics, Naval Postgraduate School, Monterey, California, April 1982.

## INITIAL DISTRIBUTION LIST

	No. Copies
1. Defense Technical Information Center Cameron Station Alexandria, Virginia 22314	2
2. Library, Code 0142 Naval Postgraduate School Monterey, California 93940	2
3. Professor R. E. Ball, Code 67Bp Department of Aeronautical Engineering Naval Postgraduate School Monterey, California 93940	2
4. LCDR R. M. Hill, USN Naval Plant Representative Office Sikorsky Aircraft Division United Technologies Corporation Stratford, Connecticut 06497	2
5. Mr. Paul Chan (220-48) Vought Corporation P.O. Box 225907 Dallas, Texas 75265	1
6. Mr. Jiten Chhabra Lockheed California Company Dept. 7525, Bldg. 67, Plant A1 P.O. Box 551 Burbank, California 91520	1
7. Mr. Walter Waskowski Northrop Corporation Aircraft Division One Northrop Avenue Hawthorne, California 90250	1
8. Capt. Steve Hollowell AFWAL/FIMB Wright-Patterson AFB, Ohio 45433	1
9. Mr. Milfred W. Brown Naval Air Systems Command Washington, DC 20361	1

	No. Copies
10. Mr. John J. Morrow (Code 338) Naval Weapons Center China Lake, California 93555	2
11. Mr. D. Bartz Sikorsky Aircraft Co. Survivability/Vulnerability Dept. North Main St. Stratford, Connecticut 06602	1
12. Mr. W. Darling Rockwell International North American Aircraft Operations P.O. Box 92098 Los Angeles, California 90090	1
13. Department Chairman, Code 67 Department of Aeronautics Naval Postgraduate School Monterey, California 93940	1
14. Combat Data Information Center ATWAL/FIES/CDIC Wright Patterson AFB, Ohio 45433	1
15. Mr. Dale Atkinson (Air-5164) Naval Air Systems Command Washington, DC 20361	1
16. Mr. Michael Meyers McDonnell Douglas Aircraft Co. P.O. Box 516 St. Louis, Missouri 63166	1

**END**

**DATE  
FILMED**

**6 - 83**

**DTIC**