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NOISE PHENOMENA IN CROSSED-FIELD ELECTRON BEAMS

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FINAL TECHNICAL REPORT

Submitted to:

Air Force Office of Scientific Research (AFSC) Building 410 Bolling Air Force Base, D.C. 20332

Prepared by:

UCA Systems Inc. 701 Welch Road Palo Alto, California 94304

Principal Investigators:

Gerald P. Kooyers	Ph.	415-328-8200
Elden K. Shaw	Ph.	415-328-8200

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1.0 INTRODUCTION

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In this Final Technical Report the results of a three years' research effort directed toward the theoretical study of noise in crossed-field beams is presented. The major findings of this study are:

- The only noise growth mechanism identified by the model in the gum region of a crossed-field beam is that due to the diocotron effect.
- (2) In order to reduce the noise arising in the gun region of a crossed-field beam, it is important that the entire cathode be operated in the temperature limited regime with focusing and other electrodes designed to give smooth, noncycloiding beams.

A more complete summary of the objectives and major accomplishments of this study is given in the next two sections. In the main body of the report, more detailed discussions of the two- and three-dimensional computer models are given together with detailed discussions of the noise simulation experiments which support the major findings given above.

1.1 <u>Research Objectives</u>

- The general research objectives as given in the contract were: The contractor shall furnish scientific effort needed to conduct the following research:
 - a. Carry out a theoretical analysis of growth of noise in cross-field microwave tubes based on solving alternately and iteratively in steps two systems of equations - one system describing the motion of each electron and the other Poisson's equation depicting the electrostatic potential corresponding to an electron concentration - with the noise identified as averages of parameters of the motion.

b. Computer aspects of (a) are to be performed employing the most efficient computer software and hardware.

1.1.1 Specific Research Tasks of the First Year's Research Effort

- Conduct a preliminary noise study using the existing twodimensional model at cathode.
- (11) Generalize model under (1) by introducing new boundary conditions and variable model size.
- (iii) Investigate two-dimensional noise models in parallel plane diodes.
- (iv) Generalize to investigate three-dimensional noise models.
- (v) Obtain quantitative effects on noise due to cathode length, magnetic field strength, cathode current density, feedbackto-potential-minimum noise mechanism, and the diocotron noise mechanism.
- (vi) Identify mechanisms which on the basis of the theoretical studies imply reduced noise levels.

1.1.2 Specific Tasks of the Second Year's Research Effort

- Implement the three-dimensional noise model, paying special attention to program efficiency.
- (ii) Initiate preliminary studies of noise mechanisms using the three-dimensional model.
- (iii) Test validity of present approximate techniques to study arbitrary electrode shapes in the two-dimensional model.
- (iv) Continue studies to obtain further quantitative data on the noise generation and noise growth mechanisms in the two-dimensional model.

(v) Investigate the extent of computer generated noise in the above tasks.

1.1.3 Specific Tasks of the Third Year's Research Effort

- (i) Perform a detailed study of noise mechanisms using the three-dimensional model.
- (ii) Improve the program efficiency of the three-dimensional model developed under this contract by coding portions of the program in assembly language.
- (iii) Include and study the effect of grids in the threedimensional model.
- (iv) Study any noise reduction techniques resulting from the study of noise mechanisms.

1.2 Summary of Research Accomplishments

The research accomplishments associated with the one- and twodimensional models are listed below. All of the specific tasks required in the first, second, and third years' effort were completed with the exception of 1.1.3 (ii), where time did not permit coding of the program in assembly language to improve program efficiency, and 1.1.3 (iii), where time constraints only made it possible to study the effects of grids in one three-dimensional gun geometry. The study of more threedimensional gun configurations with grids included would certainly have been desirable.

It should be noted that the two- and three-dimensional computer models are not only capable of giving information about the noise generated in a crossed-field beam but can also be used to analyze and design crossedfield guns in order to obtain optimum optics and other beam characteristics.

1.2.1 Research Accomplishments Associated with the Two-Dimensional Model

 An existing self-consistent two-dimensional model was successfully modified and extended to permit the study of noise in two-dimensional crossed-field geometries.

- (2) A long Kino type crossed-field gun was studied using the two-dimensional model. The steady state characteristic' of the gun agreed closely with the theoretical values predicted by the Kino gun theory.
- (3) Noise studies of the two-dimensional Kino gun led to th following conclusions:
 - Shot noise at the cathode is the most important contributor to gun noise.
 - The noise is concentrated on the upper edge of the beam as it exits the gun region.
 - Designing guns with good beam optics appears to be most important in reducing beam noise.
 - The noise increases in the gun region at an exponential rate consistent with the diocotron growth mechanism.

1.2.2 Research Accomplishments Associated with the Three-Dimensional Model

- (1) A three-dimensional computer model was developed with the following important characteristics:
 - The model is self-consistent and convergent.
 - The model includes the effects of variable magnetic fields in all three dimensions.
 - The model includes the effects of space charge.

- The model is capable of handling a variety of boundary conditions.

- The effect of grids can be taken into account in the model.
- (2) In addition to verifying the conclusions of the two-dimensional work, three-dimensional noise studies of parallel plane diode and Kino type guns have led to the following conclusions:
 - A small magnetic field perpendicular to the cathode does not decrease beam noise.
 - The smoothing effects of the space charge minimum at the cathode are present and are consistent with the results obtained by Harker.⁽¹⁾

1.3 Publications Resulting from the Work Performed

As a result of the work performed under this contract, a paper was presented at the IEDM Meeting on December 7, 1977 in Washington, D.C. A modified version of this paper was published in the IEEE Transactions on Electron Devices. Both papers are listed below.

- (1) E. K. Shaw and G. P. Kooyers, "Computer Aided Design and Analysis of Electron Guns for Injected Beam Crossed Field Amplifiers," presented at the International Electron Devices Meeting, Wash., D.C., 1977. Published in the Technical Digest of the IEDM, p. 440, 1977. (See Appendix B of this report for the complete text)
- (2) E. K. Shaw and G. P. Kooyers, "Computer Aided Design and Analysis of Electron Guns for Injected Beam Crossed Field Amplifiers," IEEE Transactions on Electron Devices, Vol. ED-26, No. 7, p. 1100, July 1979. (See Appendix C of this report for the complete text)

1.4 Principal Investigators

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 Gerald P. Kooyers, President of UCA Systems Inc., Palo Alto, California.

(2) Elden K. Shaw, Senior Research Scientist, UCA Systems Inc.,
 Palo Alto, California, and Professor of Electrical Engineering,
 San Jose State University, San Jose, California.

2.0

HISTORICAL REVIEW OF THE CROSSED-FIELD NOISE PROBLEM

In the first section below the state-of-the-art of crossed-field noise studies prior to 1977 is discussed, and the limitations and problems associated with these studies are pointed out. In the subsequent section a comparison of the studies sponsored by the Air Force Office of Scientific Research is given based on 1978-1979 annual reports and verbal communication with the other researchers.

2.1 The Problem of Noise in Crossed-Field Devices Prior to 1977

Although a considerable amount of experimental and some theoretical work was done on noise in crossed-field devices prior to 1977, many of the fundamental noise mechanisms were not well understood. Almost no quantitative results were available for predicting the amount of noise generated, and no definitive experiments had been made which showed how to reduce the noise. The justification for the above conclusions is given in Appendix A, where a review of the noise problems in crossed-field beams is given.

Ultimately almost all the noise in crossed-field beams can be traced to fluctuations in velocity, position, and emission time of the electrons emitted off the cathode surface. A study of noise thus must begin at the cathode surface. If these noise fluctuations generated at the cathode did not grow as the electrons move downstream, the problem of noise in crossed-field beams would be greatly reduced. However, as discussed in Appendix A, two mechanisms of noise growth have been identified, (2,3)and others may exist when the three-dimensional nature of the beam is considered. (4) In fact, one of these noise mechanisms (feedback to the potential minimum⁽²⁾) leads to a prediction of instability for the "long" crossed-field guns now in use. Any attempt to model and predict noise must then include a reasonable downstream portion of the beam and must not be limited to the small signal regime.

The statistical or Monte Carlo approach to the study of noise in crossed-field guns was used by Wadhwa and Rowe, ⁽⁵⁾ Pollack and Whinnery, ⁽⁶⁾ and most recently by Lele and Rowe. ⁽⁷⁾ Basically each of these approaches

is the same: the beam is approximately simulated by rods of charge (two-dimensional analyses only) which are emitted off the cathode according to the random emission processes at the cathode (with, in general, some approximations). The motion of these rods of charge is then followed in time with the motion governed by the Lorentz Force Law with the electric fields obtained from a solution of Poisson's Equation.

Unfortunately, there are two problems associated with each of these previous models which drastically limited the applicability and usefulness of the results obtained. The first problem has to do with the choice of beams studied. Only "short" guns with relatively low current density were modeled. One of the most important noise mechanisms, feedback to the potential minimum, thus was not studied and high current densities which turn out to be desirable for essentially all present CFA's were not treated at all. The second problem in each previous Monte Carlo approach has to do with computer time, and thus cost. Because the computer code for the solution of each case took a relatively long time to run, only a very limited number of cases was studied. The variation of beam noise as various parameters change is not available from these previous studies.

2.2 Comparison of AFOSR Crossed-Field Noise Studies

In Table I a comparison of the four noise studies sponsored by AFOSR is presented. This table is based on 1978-1979 reports $^{(1)}$, $^{(8)}$, $^{(9)}$ and private communications from the researchers $^{(10)}$, $^{(11)}$ involved in the studies. The comparison is not meant to be comprehensive. Rather, only the basic methods of approach and preliminary results are compared.

TABLE I

COMPARISON OF AFOSR CROSSED-FIELD NOISE STUDIES

	Shaw & Kooyers UCA Systems	Fontana MacGregor & Rowe Harris SAI	Harker & Crawford Stanford Univ.	Shkarofsky MPB Technologies
APPROACH	Lagrangian (particle)	Eulerian (fluid) Lagrangian (particle)(1)	Eulerian (fluid)	Eulerian (fluid)
Two-Dimensional Model - Completed	Yes	Yes	Yes	Yes
Three-Dimensional Model - Completed	Yes	-	No	No
Self Consistent Approach	Yes	Yes	No ⁽²⁾	No ⁽²⁾
Two-Dimensional Noise Calculations	Yes	No	Yes ⁽³⁾	Yes ⁽⁴⁾
Verified Space Charge Smoothing	Yes	-	Yes	Yes
Verified Diocotron Growth Mechanism	Yes	-	-	-
Capable of Modeling Grids	Yes	Yes	No	No
Detected Feedback to Potential Min. Instab.	No	-	No	No
Capable of Modeling Entire Gun Region	Yes	Yes	No	No
Useful as a Design Tool for CFA Guns	Yes	-	No	No
Three-Dimensional Noise Calculations	Yes	No	No	No

NOTES: (1) Fontana used a fluid approach in the two-dimensional model which gave unsatisfactory results.

(2) Both Harker and Shkarofsky assumed a parabolic d.c. potential at the cathode.

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- (3) Harker solved for current fluctuations only.
- (4) Shkarofsky solved for velocity fluctuations only.

3.0 THE TWO-DIMENSIONAL NOISE MODEL

C

In the first year's study a two-dimensional model which was originally used as a design tool for crossed-field guns was refined and modified for the study of noise in crossed-field guns. The details of this two-dimensional model are given in Appendices B and C of this report. In Fig. 3.1 a flow diagram is presented of the overall approach used in both the one- and two-dimensional studies.

The method of simulating the noisy beam at the cathode and the method of calculating noise from the beam characteristics are discussed in detail below since these topics are not discussed in detail in the Appendices.

3.1 Simulation of the Noisy Beam at the Cathode

Since, as previously stated, the ultimate source of nearly all the noise is due to random processes of electron emission at the cathode surface, it is very important that the emission properties of the cathode are modeled as accurately as possible. The cathode emission is completely specified by the following:

- (1) The number of electrons emitted in a time interval Δt
- (2) The emission velocities of the electrons
- (3) The time of emission of the electrons.

The method for calculating each of the above three quantities is summarized below.

3.1.1 Emission Number

It can be shown⁽¹²⁾ that the probability that exactly S electrons are emitted from a given cathode spot of length Δy during the time interval Δt is given by

$$f(S) = \frac{e^{-\lambda_{\lambda}S}}{S!}$$
(3-1)



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Figure 3-1. Noise Simulation and Evaluation Flow Diagram.

where λ is the average number of electrons emitted in each Δt . This is recognized to be the Poisson Distribution.

The probability that S or fewer electrons are emitted in the time interval Δt is given by the cumulative distribution function

$$F(S) = \sum_{S=0}^{S} f(S)$$
 (3-2)

and

$$0 < F(S) 1$$
 (3-3)

To determine the number of electrons emitted in a time interval, ∆t, a uniformly distributed random number, R, between 0 and 1 is generated. . The number of electrons emitted is S if

$$F(S-1) < R \notin F(S)$$

$$(3-4)$$

When $R \leq F(0)$ no electrons are emitted.

• For computational purposes, the following procedure is then followed:

- (1) The maximum number, N_m , of electrons allowed to be emitted from the area $\Delta y \Delta z$ and in any Δt is selected.
- (2) f(S) is then calculated for each value of S between0 and N_.
- (3) F(S) and F(S-1) are then calculated for each S between O and N using the calculated density functions f(S)and are then stored in a table containing S, F(S), and F(S-1).

(4) Random number R is then generated.

(5) Knowing R, the pair of values F(S-1) and F(S) that bracket R are found from the table and the corresponding value of S is then known.

Initially a value of 20 was chosen for $N_{\mathbf{R}}$. This value was used by 0'Flynn⁽¹³⁾ and Kooyers⁽¹⁴⁾ for similar problems.

3.1.2 Emission Velocities

A half-Maxwellian distribution is assumed for the x component of emission velocity. The x velocity is then given by $^{(7)}$

$$\dot{x} = \frac{2kT}{m} \left(-l_n R_x \right)^{l_x}$$
 (3-5)

where R_x is a random number uniformly distributed between 0 and 1, T is the cathode temperature, k is Boltzmann's constant, and m is the electron mass.

A full-Maxwellian distribution is assumed for electron velocities in the y direction (for the two-dimensional case) and the y and z directions (for the three-dimensional case). The relation giving these velocities is

$$\dot{y} = \frac{2kT}{m} erf^{-1}(R_y) \qquad (3-6)$$

and

$$\dot{z} = \frac{2kT}{m} erf^{-1}(R_z)$$
 (3-7)

where R and R are random numbers uniformly distributed between 0 and 1. In order to simulate these velocities efficiently, an approximate formula is used for the error functions. (15)

3.1.3 Emission Times

In order to avoid calculating the positions of the electrons above the cathode for random emission times, we assume that all S of the electrons in a given Δy are emitted at the same time but with a random distribution over the Δy . The emission points are then given by

$$y = y_0 \pm R_y \frac{\Delta y}{2}$$
 (3-8)

$$z = z_0 \pm R_z \frac{\Delta y}{2}$$
 (three-dimensional case) (3-9)

where R_y is a random number between 0 and 1 and the sign of R_y is also selected randomly. Thus, to an observer looking at one point on the cathode surface, the electrons will appear to come off that point at random times. Provided the time interval Δt is chosen to be relatively small, the degree of approximation introduced should be minor.

3.2 Noise Analysis Procedures

From the computer model we have available the positions and velocities of each electron at each discrete time $t = N_t \Delta t$ where N_t is the iteration number. In order to be useful, this information must be related to the noise. There are several different noise-related parameters which are calculated in this study.

In general we would like to be able to calculate each of these parameters at any arbitrary point (x, y, z) within the beam. However, the nature of the way we solve the problem using a finite number of electrons limits us to finding the fluctuations in some small but finite region. In order to obtain enough "electrons" in each region for reasonably good statistics, the gun region is divided into 512 noise regions which are 3 x 3 mesh units in size. These regions are sketched in Figure 3-2. The method used for calculating each of the parameters of interest is discussed below.



3.2.1 A.C. Velocity Components

In order to obtain the a.c. velocity components we first must find the average velocity. The a.c. velocity is then equal to the actual velocity minus the average velocity. The velocity averaged over one noise region at a given time t is given by

$$\langle v_{x_{i,j}}(t) \rangle = \frac{1}{N_{i,j}} \sum_{\ell=1}^{N_{i,j}} v_{x_{i,j}}(t,\ell)$$
 (3-10)

where $v_{x_{i,j}}$ (t, l) is the x velocity of the lth electron in the i,j i,j-th noise region. Since the equations for the other component v_{y} are of the same form, we carry the analysis only for the v_{x} component. Over the time interval $T = N_{T} \Delta t$ the time average velocity is given by

$$< \overline{\mathbf{v}_{\mathbf{x}_{i,j}}} = \frac{1}{N_T} \sum_{m=0}^{N_T} \frac{1}{N_{i,j}} \sum_{\ell=0}^{N_{i,j}} \mathbf{v}_{\mathbf{x}_{i,j}}(t_m,\ell)$$
 (3-11)

The a.c. velocity component can now be calculated at time t from

$$\tilde{\mathbf{v}}_{\mathbf{x}_{i,j}}(t) = \langle \mathbf{v}_{\mathbf{x}_{i,j}}(t) \rangle - \langle \overline{\mathbf{v}_{\mathbf{x}_{i,j,k}}} \rangle$$
 (3-12)

where \bar{v} (t) is the a.c. component of velocity in the i,j-th noise $x_{i,j}$ region at time t.

3.2.2 Variance of the Velocity Components

The variance of the x-velocity at time t can be found from

$$v_{\mathbf{x}_{i,j}}^{2}(t) = \langle v_{\mathbf{x}_{i,j}}^{2}(t) \rangle - \langle v_{\mathbf{x}_{i,j}}(t) \rangle^{2}$$
 (3-13)

where $\sigma_{x_{i,j}}^2$ (t) is the variance of the x-component of velocity in the i,j i,j-th noise region at time t and $\langle v_{x_{i,j}}^2$ (t)> is given by

$$\langle v_{x_{i,j}}^{2}(t) \rangle = \frac{1}{N_{i,j}} \sum_{\ell=1}^{N_{i,j}} v_{x_{i,j}}^{2}(t,\ell)$$
 (3-14)

The variance of the other component of velocity can be found in a similar manner. The variance and the a.c. components of velocity are useful in that they give the relative magnitude of the fluctuations. Thus the "growth" of these fluctuations can be studied by calculating the a.c. velocity components and variances at various downstream positions in the beam. The a.c. components of velocity and their variance do not give any information about the frequency spectrum of these components.

3.2.3 A.C. Current Components

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The charge density in the i,j-th noise region at time t is given by

$$P_{i,j}(t) = \frac{-|e| N_Q N_{i,j}}{\Delta x \Delta y}$$
(3-15)

where N_Q is the scaling factor for the electrons. The time averaged space charge density in the i,j-th noise region is given by

$$\langle \rho_{i,j} \rangle = \frac{-|e| N_0}{\Delta x \Delta y \Delta z} \frac{1}{N_T} \sum_{m=0}^{N_T} N_{i,j}(t_m)$$
 (3-16)

Now the current density at time t in the i, j-th rectangle is given by

$$J_{x_{i,j}}(t) = \rho_{i,j}(t) \dot{x}_{i,j}(t)$$
(3-17)

and the time average current density is given by

$$\langle J_{x_{i,j}} \rangle = \langle \rho_{i,j} \rangle \cdot \langle v_{x_{i,j}} \rangle$$
(3-18)

The a.c. component of the x-component of current density is thus given by

$$\tilde{J}_{x_{i,j}}(t) = J_{x_{i,j}}(t) - \langle J_{x_{i,j}} \rangle$$
 (3-19)

Again, although this parameter does not tell us anything about the frequency components of the fluctuations, it is very useful in determining the relative growth rate of current fluctuations in the beam.

3.2.4 Correlation Functions and Power Spectra

Finally, in order to completely determine the noise properties of the beam, it is necessary to calculate the relative magnitudes of the various frequency components of the noise at a particular crosssection of the beam. The two quantities of interest in beginning this calculation are the a.c. kinetic potential and the a.c. current fluctuation. These two quantities may be defined for a two-dimensional beam as⁽⁷⁾

$$\tilde{v}(t) = \frac{-\bar{\mu}_0 \cdot \bar{v}(t)}{n}$$
(3-20)

and

$$\tilde{J}(t) = \frac{\tilde{J}(t) \cdot \tilde{J}_0}{|\tilde{J}_0|} - J_0$$
 (3-21)

where V(t) is the a.c. kinetic potential at time t

 $\vec{\mu}_0$ is the d.c. velocity of the beam $\vec{\nu}(t)$ is the a.c. component of the velocity of the beam η is the charge-to-mass ratio of the electron $\vec{J}(t)$ is the a.c. current density fluctuation at time t $\vec{J}(t)$ is the a.c. current density in the beam \vec{J}_0 is the d.c. current density in the beam. At this point the final problem in the spectral analysis is in relating a spectrum consisting of finite discrete data of a function $\tilde{V}(t)$ or $\tilde{J}(t)$ (each value obtained as an average over a small interval Δt) to the true spectrum of the continuous time function. Several authors have treated this subject in some detail. Results show that it is possible to calculate the Haus noise parameters $\phi(f)$, $\psi(f)$, and $\theta(f)$ from the discrete data where

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- $\phi(\omega)$ is the self power density spectrum of the kinetic noise voltage,
- $\psi(\omega)$ is the self power density spectrum of the noise current modulation, and
- $\theta(\omega)$ is the cross-power-density spectrum between the kinetic voltage and the current modulation.

For a crossed-field beam an exact relation between the noise parameters and the noise figure has not been derived. However, even without an exact expression, a study of the relative magnitudes of the various noise parameters (ϕ , ψ , and θ) as a function of position and other gun and beam parameters is very valuable in gaining an understanding of the noise mechanisms in crossed-field electron devices.

4.0 TWO-DIMENSIONAL NOISE SIMULATION EXPERIMENTS

Using the two-dimensional model, several noise simulation experiments have been carried out on a long Kino gun. The geometry and parameters of the long Kino gun are given in Appendix C. In particular, the noise growth rate, the variation of noise throughout the gun region, and the effect of changes in the beam injection conditions have been studied.

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4.1 Noise Growth in the Gun Region

In Figure 4-1 the noise power spectral density due to fluctuations in the kinetic potential is plotted from the end of the cathode to where the beam exits the gun region. The noise growth appears to be approximately linear when plotted on semi-log paper, and thus the actual noise growth is exponential. For the case of Figure 4-1 a straight line approximation gives a noise growth of approximately 170 db/in which is a rather substantial growth rate. The diocotron theory also predicts an exponential growth of the noise.

4.2 Noise Variation Across the Beam

Shown in Figure 4-2 and Figure 4-3 is the variation in noise from the bottom to the top of the beam as the beam exits the gun region.

In all of these cases, noise is greater on the top edge of the beam then on the bottom edge. This may be due to the fact that the electrons on the top edge of the beam have traveled a greater distance and thus have experienced more diocotron noise growth than those electrons on the bottom edge of the beam.

Unfortunately, as the beam enters the interaction region of a CFA, the noisiest part of the beam will be nearest the slow wave circuit and thus will be more strongly coupled to the slow wave circuit.



Figure 4-1. Noise Variation with Distance for Two Cathode Temperatures.





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4.3 Variation of Noise With Frequency

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Shown in Figure 4-4 and Figure 4-5 is the variation in beam noise as a function of frequency for two different positions in the beam. For Figure 4-4 the noise is sampled near the cathode; while in Figure 4-5 the noise is plotted for a noise region near the gum exit.

At the gun exit, the noise as measured by the self power spectral density, ϕ , of the kinetic potential is fairly constant with frequency, while the self power spectral density of the current density, ψ , has large peaks and valleys which roughly correspond to multiples of the cyclotron frequency.

Near the cathode both ϕ and ψ vary with frequency and have roughly the same valleys and peaks corresponding to multiples of the cyclotron frequency. The variation of noise with frequency near the cathode would be unexpected if it were not for the fact that many of the electrons near the cathode have been in the gun for a relatively long time due to two effects:

- Electrons returning to the gun on a cycloidal path from further upstream
- (2) Electrons which migrate downstream while maintaining a position very close to the cathode.

In both cases noise growth mechanisms have sufficient time to affect the noise characteristics of these electrons.

4.4 Variation of Noise with Cathode Entrance Conditions

In order to study the relative effects of the various noise generation mechanisms at the cathode, computer runs were made with the following injection conditions:



Fig. 4-4. Self Power Spectral Densities for Current and Kinetic Potential Fluctuations Near the Cathode.



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(1) Random positions and random numbers* injected but with four different injection velocities:

- (a) Zero velocity (Temperature = 0)
- (b) Uniform velocity

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- (c) Maxwellian velocities (Temperature = T)
- (d) Velocities consistent with the electric fields at the cathode.
- * But with constant <u>average</u> current emitted from the cathode surface and with the cathode space charge limited.
- (2) Random positions, random numbers injected, and Maxwellian velocities but with variable amounts of charge per "electron" from the cathode surface.

Noise outputs for the various input conditions of cases (1) above indicate that the <u>noise is not appreciably affected by the electron injection</u> <u>velocities</u>. A typical plot of noise output for simulated cathode temperatures of 0 and 1323°K is given in Figure 4-1.

In all the cases (1) above, the cathode is running in a spacecharge-limited condition. If the numbers of electrons emitted from the cathode are reduced to the point where the cathode becomes temperature limited, the optics of the gun deteriorate and the noise in the beam increases dramatically.

When the total charge carried by one "electron" is increased, the shot noise is increased in direct proportion. This corresponds to the input condition of cases (2) above. Noise output from the computer simulation as measured by the velocity variance also goes up in direct proportion to the amount of charge per "electron." <u>These results indicate</u> that shot noise is the main contributor to beam noise in this model.
5.0 THE THREE-DIMENSIONAL NOISE MODEL

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The three-dimensional noise model solves for the motion of charges in a three-dimensional rectangular geometry including the effects of static electric and magnetic fields and space charge. A computer program written in FORTRAN IV carries out the steps given in the flow chart of Figure 5-1.

A sample of the output generated by this program is given in Appendix F. Information is printed out giving cathode current density, distribution of charges in the beam, selected charge trajectory information, etc. for each time step.

Below, details are given on the potential, trajectory, and noise calculations.

The three-dimensional program consists of the major program modules shown in Figure 5-2. The function of each program module is given below:

- GUN3D Main calling program which reads in gun data, initializes program variables, and calls other program modules.
- TRAJ3D Solves the three-dimensional trajectory equations including variable magnetic fields in the x, y, and z directions.
- EMIT3D Calculates particle emission numbers, positions, and velocities as they are emitted from the cathode.
- POSS3D Solves Poisson's Equation in three dimensions using subroutines FOUR67 and PWSCRT.
- NOIS3D Calculates current and velocity fluctuations in the beam and uses these fluctuations to calculate the space charge smoothing factor and the "temperature" of the beam at various locations within the beam.



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Solving Poisson's Equation in Three Dimensions

In order to obtain the potentials in the three-dimensional region, it is necessary to solve the Poisson Equation

$$\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} + \frac{\partial^2 \phi}{\partial z^2} = -\rho/\epsilon_0 \qquad (5-1)$$

in the region sketched in Figure 5-3. The boundary conditions shown in Figure 5-3 are chosen to be consistent with three-dimensional crossed-field gun geometries. Boundary conditions on the z-boundaries allow for potentials with the derivative equal to zero, periodic potentials, or constant potentials.

The finite difference techniques used in the solution are given in detail in Appendix D for the case of fixed potentials on the z-boundaries and follow a method suggested by Swartrauber and Sweet. (16) Figure 5-4 shows the mesh geometry used in the finite difference solution. A very brief review of the technique is given below.

> The finite Fourier expansion of the five point difference approximation to Poisson's Equation is taken in the z-direction using a program which performs the Fourier transform⁽¹⁷⁾ (FOUR67). This leads to n-1 <u>two-dimensional</u> Helmholtz equations

$$\nabla^2 \overline{\phi} + K^2 \overline{\phi} = \overline{\rho} \tag{5-2}$$

where $\overline{\phi}$ and $\overline{\rho}$ are the transformed potentials and space charge and K is a constant which depends on n. The space charge assigned to each mesh point is found by adding together the total number of particles in a cell $(h_x \times h_y \times h_z)$ centered on the mesh point. This is commonly called the particle in cell (PIC) approach.

(2) Equation (5-2) is then solved for the transformed potentials using an existing program (PWSCRT). (16)



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Figure 5-4. Three-Dimensional Geometry with Modified Boundary Conditions.

(3) Given the transformed potentials a program (FOUR67) (17)
 is used to obtain the actual potentials by performing the inverse Fourier Transform.

5.2 Equations of Motion

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A detailed discussion of the development of the three-dimensional finite difference equations of motion is given in Appendix E. There are several facts which should be noted concerning these equations:

- The finite difference equations of motion are accurate to second order and are solved in a right-handed coordinate system.
- (2) The electric field is assumed to be a slowly varying function of position (i.e., does not change much in one time step).
- (3) The magnetic field is assumed to consist of x, y, and z components which may be slowly varying functions of position.
- (4) The equations are solved in such a way as to be <u>exact</u> for <u>any</u> size time step provided the electric and magnetic fields are constant.

5.3 <u>Noise Analysis Procedures</u>

A separate noise analysis program (NOIS3D) has been written to perform noise analysis on the three-dimensional beam. This program calculates two parameters which are closely related to beam noise: the space charge smoothing parameter (current fluctuations) and the beam temperature (velocity fluctuations). Details of the techniques used to make these calculations are given below.

5.3.1 <u>Current Fluctuations</u>

In order to determine the growth of current fluctuations in the beam, the current fluctuations at the cathode are calculated using computer generated values of instantaneous cathode current, I_{K} , and a running average of the cathode current, $<I_{K}>$. At each iteration, the current fluctuation at the cathode is calculated from

$$\tilde{I}_{K} = I_{K} - \langle I_{K} \rangle$$
(5-3)

Similarly the fluctuations in the beam current at various planes in the gun region are calculated from

$$\tilde{I}_{B} = I_{B} - \langle I_{B} \rangle$$
(5-4)

at each iteration.

A space charge smoothing factor, Γ , is then calculated by taking the r.m.s. of the ratio of the current fluctuations for the beam at a given plane within the gun to the cathode current fluctuations. The space charge smoothing factor is then defined to be

$$\Gamma = \left(\frac{\langle \tilde{I}_{B}^{2} \rangle}{\langle \tilde{I}_{K}^{2} \rangle}\right)^{1/2}$$
(5-5)

5.3.2 Velocity Fluctuations

If the beam can be considered to be in a state of equilibrium with a Maxwellian distribution of velocities, then the temperature of the beam may be calculated from

$$T = \frac{m}{3k} < (v - \langle v \rangle)^2 >$$
 (5-6)

where m is the electron mass, k is Boltzman's constant, and v is the instantaneous velocity. In terms of the components of velocity, the above equation may be written as

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$$T = \frac{\pi}{3k} \left(\langle \tilde{v}_{x}^{2} \rangle + \langle \tilde{v}_{y}^{2} \rangle + \langle \tilde{v}_{z}^{2} \rangle - \langle \tilde{v}_{x}^{2} \rangle^{2} - \langle \tilde{v}_{y}^{2} \rangle^{2} - \langle \tilde{v}_{z}^{2} \rangle^{2} \right)$$
(5-7)

where the brackets, < >, represent spatial averages in the region under analysis. Running averages (time averages) of the x,y,z velocities in each region under analysis are calculated at each time step and used to determine the instantaneous velocity fluctuations.

The temperature, T, is calculated in rectangular regions one mesh distance thick in the xy-plane, the xz-plane, and the yz-plane as sketched in Figure 5-5. It is thus possible to compare the beam temperature at various x, y, and z planes.

5.4 Inclusion of Electrodes (Grids) in the Model

Two approaches to the inclusion of the grids in the three-dimensional geometry are discussed below:

5.4.1 Approximate Technique Using D.C. Potentials

In this approach the principle of superposition is used to include internal electrodes in an approximate way as follows:

 Using the exact geometry of the gun including grids and other internal electrodes, the potential is solved in the three-dimensional geometry neglecting space charge. In the discussion below these potentials are called the d.c. potentials.



(2) Poisson's Equation is solved for the potentials in a rectangular "box" (with zero potential on the electrodes) which approximates the exact gun geometry as closely as possible and the cathode exactly.

- (3) The d.c. potentials are added to the space charge potentials to obtain the potentials used in the trajectory equations.
- (4) Steps (2) and (3) are repeated for each iteration. Note that the d.c. potentials need only be calculated once and then saved.

A three-dimensional relaxation program for solving Laplace's Equation with arbitrary internal electrodes and boundary conditions has been written. The program has been tested on a three-dimensional crossedfield diode with grids in a 16 x 16 x 16 mesh convergence point system. After 50 iterations (.08 minutes CPU time) the changes in potential are less than 1% and the program appears to have converged.

This approach is approximate in that the imaging of the electrons in the grid are neglected. Since the grids are normally run at a negative potential, this error is expected to be relatively small.

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5.4.2 <u>Exact Approach Using a Relaxation Procedure on the Space Charge</u> Hockney⁽¹⁸⁾ has suggested several methods of including the effect

of electrodes in the interior of the mesh region used for the solution of Poisson's Equation. When the number of electrode points is large, he suggests that the space charge on individual mesh points be adjusted iteratively in response to the local error in potential on that point. This approach has been programmed as a test case in the Poisson solving routines. The procedure used is outlined below: Storage in an array is provided for the coordinates, potential, and space charge of each internal electrode point.

- (2) An initial value of charge is assigned to each internal electrode point assuming the electrode is an isolated sphere of dimension $h_x X h_y X h_z$.
- (3) After each iteration of Poisson's Equation a new value of charge is assigned to each electrode point according to the following relaxation formula

$$Q_{new} = Q_{old} \left[1 + R_Q \left(\frac{V_{old} - V_0}{|V_{max}|} \right) \right]$$

where R_Q is a relaxation factor ≤ 1 , V_Q is the electrode potential desired, and V_{max} is the maximum of V_{old} and V_Q .

The above technique has been tried on a parallel plane geometry test case with one interior electrode point at -100 volts. The procedure was found to converge with a relaxation factor of .25. Time did not permit the inclusion of this technique in the study of a gridded gun. Obviously, this technique will result in an increase in run-time for the computer program. However, the increase in run-time may be as small as a factor of two since the technique can be applied without a time penalty during the iterations necessary for the solution to reach a steady state. Once the steady state is reached the correction to the charges necessary to maintain the correct internal electrode potential should be small and thus easily accomplished with a few iterations.

6.0 THREE-DIMENSIONAL NOISE SIMULATION STUDIES

Using the three-dimensional model, several test cases were run to determine the correctness of the model and obtain preliminary noise information.

In testing the model for accuracy of potential and trajectory calculations, a 16 x 16 x 16 mesh was used in the parallel plane geometry of Fig. 6-1. For the noise calculations, a 32 x 48 x 16 mesh was used. For the larger mesh system with 16,000 charges, each case required approximately 12 minutes of computer time on the IEM 370/168. The results obtained using the larger mesh are discussed below.

6.1 The Long Kino Gun Test Case

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As a preliminary test case, the Long Kino Gun sketched in Figure 6-1 was approximately modeled in three dimensions using a $32 \times 48 \times 16$ mesh system. The z-boundaries are taken to be at zero potential in order to confine the beam. The potential on the lower boundary (cathode) is taken as zero potential with the potential on the anode fixed by the potentials from the Kino gun theory. This gun has the following design parameters:

Total Beam Current	3.34 amps
Cathode Length	.150 inches
Cathode Width	.247 inches
Magnetic Field	2500 gauss
Current Density	14 amp/cm^2
Accelerator Potential	5000 volts
Normalized Cathode Length	116.4 Kino units

Figure 6-2 shows the convergence characteristics of the computer simulation. In this figure the cathode current reaches a "steady state" in about 10 iterations and then fluctuates about this "steady state" for time steps 10 through 75. The computer calculated average exiting beam current is 3.62 amps which is near the design value.









Figure 6-2. Cathode Current in the Long Kino Gun as a Function of Iteration Number.

6.1.1 <u>Current Fluctuation Calculations</u>

Shown in Figure 6-3 is the space charge smoothing factor, Γ , as a function of iteration number for the Long Kino Gun test case with the cathode operating space charge and temperature limited. In the space charge limited case, the current fluctuations are smaller at the beam exit than at the cathode surface. This indicates space charge "smoothing" by the potential minimum and is consistent with the results obtained by Harker. (1) In the temperature limited case the current fluctuations at beam exit are larger than at the cathode surface. In an ideal case of no noise growth mechanisms, the value of Γ for the temperature limited case would be unity. The value of $\Gamma = 1.75$ is due to diocotron noise growth in the beam as it travels from the cathode to the beam exit plane. Space charge smoothing is shown even more clearly in Figure 6-4 where the space charge smoothing factor, Γ^2 , is plotted as a function of the relative current emitted off the cathode. The value of Γ^2 is calculated at a position above the cathode where Γ^2 is minimum. As expected, the space charge smoothing factor is approximately unity in the temperature limited regime and drops dramatically as the cathode is operated space charge limited. Two test cases were run with a small x-directed magnetic field. The value of Γ after 75 iterations are given below:

Case I - No magnetic field in x-direction $B_{z} = .25 \text{ w/m}^{2}$ $\Gamma = .160$ $Case II - B_{z} = 52 B_{z}$ $B_{z} = .25 \text{ w/m}^{2}$ $\Gamma = .2005$

Case III - B = 10% B

$$B_z = .25 \text{ w/m}^2$$

 $\Gamma = .1883$.

In both Cases I and III the x-directed magnetic field increases the exiting current fluctuations.





6.1.2 Beam Temperature Calculations

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Shown in Figures 6-5, 6-6, and 6-7 is the temperature of the beam as a function of noise region in the x, y, and z planes. Figure 6-5 shows how the beam temperature increases toward the beam exit. This growth appears to be nearly exponential over the cathode surface, drops an order of magnitude at the end of the cathode, and then rises again at the same exponential rate. This exponential growth seems to be due to diocotron noise growth.

Temperature variations as a function of position above the cathode are shown in Figure 6-6. As found in the two-dimensional studies, the top of the beam seems to be noisier than the bottom. This could be due to diocotron growth variations since the electrons on the beam top travel further than those on the bottom.

In Figure 6-7 the variation of beam temperature is given in the direction of the magnetic field for the case of no magnetic field in the x-direction and a small magnetic field in the x-direction. Note that the magnetic field increases the temperature of the beam.

6.2 Three-Dimensional Studies With Grids

As a preliminary test case a three-dimension parallel plane diode was studied using a 16 x 16 x 16 mesh system. The diode had a single grid at -100 volts running the full length of the cathode surface. The grid was simulated using the technique of para. 5.4.1 where the dc potentials are added to the space charge potentials to obtain the total potential for each iteration.

In order to test the effect of grids on the noise in a crossed-field gun, the gridded injected beam gun used by Northrup and discussed by Fontana⁽⁹⁾ was simulated using a 48 x 24 x 16 mesh system. Although the calculation converged in approximately 50 iterations an error in the dc potentials was discovered which prevented the beam from entering the interaction region. Unfortunately time did not permit further calculations on this test case.





Figure 6-5. Beam Temperature in the x-z Plane.





7.0 <u>ACCURACY AND ERROR CONSIDERATIONS IN COMPUTER SIMULATIONS</u> The sources of error due to the computer formulation have been identified. These errors can be divided into three categories:

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(1) <u>Truncation errors</u> due to the finite difference approximations used to represent the Lorentz Force Equation (trajectory equations) and Poisson's Equation (potential equation). Such errors are dependent upon the magnitude of the mesh intervals in both time and space.⁽¹⁹⁾ In order to minimize these errors, the finite difference equations have been carried out to second order. The convergence of the calculations indicates that the finite difference procedure is stable.

- (2) <u>Round-off errors</u> due to the inexact representation of floating point numbers in the computer. The error depends on how a number is "rounded off" in the lowest "bits" of a 'word.' The three-dimensional model is solved on an IEM 370/168 using single precision floating point words which have an accuracy of 6 or 7 decimal digits. Clearly any fluctuations in parameters which are approximately 10⁻⁶ below the average or dc value cannot be calculated accurately. Both the velocity and current fluctuations obtained in the model are approximately two orders of magnitude larger than the round-off values. '
- (3) <u>Errors intrinsic to the simulation</u> which arise from the following approximations:
 - (a) Representation of individual electrons by a point charge (big electron) containing many electrons. This approximation results in an exaggerated collision frequency for the electrons.

(b) Representation of the space charge assigned to a particular mesh point by a uniform cloud centered on a mesh point or by some other method. This error has been investigated by Hockney⁽²⁰⁾ and Birdsall.⁽²²⁾

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(c) Representation of the functions by finite space and time steps which lead to a lack of energy conservation in the model and thus a "heating" of the electrons. This problem has been investigated by Hockney.

The magnitude of the errors caused by (a), (b), and (c) above are difficult to estimate. Hockney ^(21,22) has investigated various techniques which reduce these errors. Unfortunately, the techniques suggested by Hockney all require increased computer time and thus increase the cost of running the program. One reasonably simple technique called the cloud-in-cell (CIC) approach has proved to be effective in reducing the noise due to (b) and (c) above. Unfortunately, time did not permit the implementation of the cloud-in-cell approach.

(4) <u>Errors in the model</u> due to the approximation of the exact electrode shapes with planar electrodes.

In the two-dimensional model sketched in Figure 7-1 the space charge contribution to the electric field is found using the idealized rectangular geometry shown by the dotted lines. This results in errors since some of the charges are not imaged in an electrode at the proper distance from the charge.

An estimate of this error can be found by solving for the electric field due to a line charge imaged in a perfect conductor as sketched in Figure 7-2. For this case, the electric field is given by

$$\mathbf{E}_{\mathbf{x}_{0}} = \frac{\mathbf{q}}{2\pi\epsilon_{0}} \left(\frac{1}{\mathbf{x}-\mathbf{d}} - \frac{1}{\mathbf{x}+\mathbf{d}}\right)$$
(7-1)







where q is the line charge and d is the distance between the line charge and the conductor. Note that the second term is due to the conductor and reduces the field below that of a "free" line charge.

If d is in error by an amount Δd the above equation becomes

$$\mathbf{E}_{\mathbf{x}_{1}} = \frac{\mathbf{q}}{2\pi\varepsilon_{0}} \qquad \frac{1}{\mathbf{x}-\mathbf{d}-\Delta\mathbf{d}} - \frac{1}{\mathbf{x}+\mathbf{d}+\Delta\mathbf{d}}$$
(7-2)

The percent error due to the incorrect position of the conductor is given by

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Electric Field Error =
$$\frac{\frac{E_x - E_x}{1} x_0}{\frac{E_x}{0}} \times 100\%$$
 (7-3)

Using Equation (7-1) and (7-2) in Equation (7-3) we obtain

Electric Field Error =
$$\frac{r/d \Delta}{r/d+2(1+\Delta)} \times 100Z$$

where r = x-d, the distance from the line charge to the field point, is kept fixed and $\Delta = \Delta d/d$. This equation is plotted in Figure 7-3 as a function of r/d for various values of Δ as parameters.

We note from Figure 7-3 that as long as the field point is relatively close to the charge (r/d << 1), the error in the value of the electric field is small even for large errors in the position of the conducting plane $(\Delta \sim 1)$.

For the gun of Figure 7-1 the top of the beam is approximately 20 mesh units from the exact electrode and 28 mesh units from the approximate electrode. This gives $\Lambda = .4$. The beam thickness is approximately 15 mesh units. Thus, at the middle of the beam r/d = .33 and the maximum error in the electric field at the center of the beam would be only 4.27. In fact, the error in the total field at the center of the beam would be



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even less since the field at the center of the beam would be dominated by near charges rather than those at the edge of the beam.

A larger error would occur at the bottom of the beam where at exit the beam is approximately 6 mesh units from the exact electrode and 11 mesh units from the approximate electrode. This gives a value of $\Delta = .833$ and at the center of the beam r/d * 1.2. From Figure 7-3 the error is approximately 20.5%.

Again, the actual error would be smaller due to the shielding effects of the nearer charges.

It should be noted that in the critical region of the cathode surface, the idealized and exact gun geometries coincide so there is negligible error in the calculation of electric field near the cathode surface. The field due to charges is not the total field since there is a potential applied between the accelerator and cathode electrodes. This would further reduce the net error in the electric fields. 8.0 CONCLUSIONS AND RECOMMENDATIONS

This study shows that it is possible to model three-dimensional crossed-field beams using computer simulations and to obtain useful information about the beam characteristics from the models.

Further studies of the three-dimensional beams with grids over the cathode would be useful in order to determine the effect of grids on the beam noise and optics.

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APPENDIX A

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A SURVEY OF NOISE IN CROSSED-FIELD BEAMS

A.1 Introduction

This survey of papers on noise and noise mechanisms in crossedfield electron beams is not intended to be comprehensive, but rather is intended to be definitive in establishing the current state of knowledge concerning noise mechanisms in crossed-field beams.

Three noise mechanisms have been proposed to account for the noise growth in crossed-field beams: diocotron gain, feedback to the potential minimum and noise enhancement due to electron travel along magnetic field lines (three-dimensional effects). Each of these mechanisms is discussed below, and an effort is made to establish current knowledge concerning each mechanism, noise measurements related to each mechanism and possible methods of noise reduction.

A.2 Diocotron Noise Growth

Below, theoretical calculations, experimental measurements and possible methods of reducing diocotron noise growth are cited from the svailable literature. It should be noted that accurate and definitive quantitative measurements are difficult to make, which explains in part the lack of experimental measurements. It should also be noted that the theoretical calculations cited are only approximate since the complexity of the problem makes it very difficult to obtain closed form solutions even in the small signal regime where most of the analyses apply. One other problem with the theoretical and experimental results which should be cited is the almost total lack of noise data on high current density "long" guns since most of the experimental laboratory work was done on low current density beams.

A.2.1 Theoretical Calculations and Experimental Results

In an analysis of wave propagation in slipping streams of electrons, MacFarlane and Hay (1949) found that fluctuations in a stream of electrons will grow even without the presence of a slow wave circuit. This effect in crossed-field beams has come to be known as the "diocotron" effect. Gould (1957) developed an equation for the growth rate of signals due to this effect between parallel conducting planes for both centered and non-centered beams. Later, Sasaki and Van Duzer (1966) used coupled mode theory to obtain essentially the same result as Gould. The results of Gould and Sasaki and Van Duzer were limited in that only the three space charge waves were used in the analysis, whereas, in general, there are five possible waves which can propagate on a crossed-field beam. Finally, Mantena (1968) considered all five waves and developed a general transfunction matrix which gives the diocotron growth for a thin, non-accelerating beam in a drift region between parallel conducting boundaries. Unfortunately none of the above results can be applied to the gun region since the beam is accelerating and the electrodes in the gun region are not, in general, parallel. However, numerous experiments have confirmed that the growth of waves due to the diocotron effect does take place and the theoretical results for thin, non-accelerating beams in the small signal regime agree closely with experiment.

Van Duzer (1961) was the first one to try to calculate the growth of fluctuations in the gun region. Using modified Liewellyn-Peterson equations he was able to approximately calculate the growth

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rate of velocity, current, and position fluctuations along the beam. Later, Van Duzer (1963) used these results to try to calculate the noise figure of a crossed field amplifier including a gun region, a drift region and an amplifying region. His calculated values were very much higher than those measured on an amplifier operating at a frequency of 1874 MHz. Using an eigenvalue perturbation technique together with Monte Carlo calculations Wadhwa (1964) concluded that the growth of fluctuations in the beam is mainly due to diocotron growth during the latter part of the beam trajectory in the gun region. He also cited some experimental work which seems to confirm the above conclusion. Using the results of his calculations in 1964 for the gun region Wadhwa (1968) calculated the noise figure of an injected beam CFA including both a three and five wave analysis. A comparison of his results with those of Mantens and Van Duzer (1963) shows good correlation between experiment and theory.

A.2.2 Reduction of Diocotron Noise Growth

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In order to reduce the growth of fluctuations due to the diocotron effect, various schemes have been proposed. These techniques are reviewed below together with an attempt to evaluate the usefulness of each approach.

A.2.2.1 Modification of beam and geometry design parameters

By examining the gain equation for the diocotron growth, Wadhwa (1964) proposed several design conditions which would reduce the diocotron growth. A low current density, high velocity, high
magnetic field beam would help reduce the diocotron growth. Unfortunately, for practically all of the applications now envisioned for CFA's relatively low velocity, high current density beams are desired with magnetic fields as small as possible in order to reduce the weight of the device.

Two other suggestions of Wadhwa do have merit. First, he proposed eliminating the length of the transition region between the gun and the interaction region. This has been done in all current CFA designs and although no careful quantitative measurements have been made to verify that the noise is reduced, qualitative results would indicate that it is.

Second, he proposed that the beam be positioned asymmetrically between the focusing electrodes in the gun and transition regions. This is a design parameter which is somewhat under the control of the designer.

A.2.2.2 Noise transformer

Wadhwa and Rowe (1963) proposed that a noise transformer somewhat similar to that used in O-type tubes might be useful in reducing the noise output of a crossed field gun. Wadhwa and Van Duzer (1965, 1968) used a noise transformer in an experimental S-band CFA and achieved a 3.5 db noise figure. They attribute the low noise figure obtained to be due, in part, to the noise transformer. Sidhu and Wadhwa (1970) proposed a noise transformer composed of two parallel plate regions and suggest that such a device between the gun and the interaction space would reduce the noise as well as allowing wide dynamic range operation of the gun. Van Duzer and Harris (1964) found an instability in trajectory calculations they were attempting to make on a long Kino Gun and also proposed a noise mechanism caused by feedback to the potential minimum to account for this space charge instability. Davies and Polter (1965) made noise measurements on a long Kino Gun and were able to identify two types of noise-diocotron noise and noise which could be associated with feedback to the potential minimum. Nunn and Smol (1964) also found that the noise in crossed-field guns increases with the length of the cathode.

Ho and Van Duzer (1968) proposed a feedback model for the potential minimum region of a crossed-field gun which gave at least qualitative agreement with the experimental results cited above. They developed a stability criteria for avoiding excess noise due to feedback to potential minimum of

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where l_k is the normalized cathode length and l_c is the normalized length of a cycloid in the gun region.

It seems relatively certain that the mechanism of feedback to the potential minimum is causing excess noise in crossed-field guns even though very few quantitative measurements have been made and the theory due to Ho and Van Duzer is only approximate and limited to the small signal regime.

A.2.2.3 Shielded cathode

Theoretical work by Buneman, Levy, and Linson (1966) indicates that a partially shielded gun (shielded from the magnetic field) may have lower diocotron growth than a gun where the magnetic field does not lace the cathode. No experimental work has been reported in the area of shielded cathodes.

A.2.2.4 Non-equipotential cathodes

Pötzl (1966) has done some theoretical work on non-equipotential cathodes which is interesting from a theoretical, if not practical, point of view. His calculations indicate that if it were possible to build a non-equipotential cathode, a more stable beam would result with decreased diocotron noise growth. No experimental work has been attempted in the area of non-equipotential cathodes.

A.3 Feedback to the Potential Minimum

A.3.1 Experimental Measurements and Theoretical Calculations

As crossed-field guns were made "longer" in order to give higher current density beams, certain anamolous effects were observed. Epsztein (1957) noticed that large values of sole current occurred at certain values of the magnetic field in "long" guns. Arnaud (1964) was able to relate these results to the ratio of the height of the beam above the cathode to the length of the cathode and proposed that feedback to the potential minimum could be the cause of the excess noise. A.3.2 Reduction of Noise Growth Due to Feedback to the Potential Minimum

Since feedback to the potential minimum is a space charge effect any scheme or method which reduced the space charge in the gun region should also reduce the noise due to feedback. Significant reductions in noise were observed by Arnaud and Doehler (1961) and Arnaud, Diamond, and Epsztein (1962) when a grid was added above the gun with the grid wires perpendicular to the direction of the magnetic field and the grid close to the cathode surface. One possible explanation of the reduction in noise due to the grid is the reduction of space charge in the potential minimum (fields from the beam terminating on the grid and thus reducing the space charge fields) and thus a reduction in the feedback noise mechanism. Such a conclusion must be considered tentative since no experimental or theoretical work has been done to determine why or how a grid reduces the noise.

A.4 Three-Dimensional Effects

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All of the theoretical analyses which have been made of crossedfield beams have been two dimensional. Variations and fluctuations of the beam in the magnetic field direction have been ignored. Gandhi (1965) made some calculations which indicated that random emission velocities in the direction of the magnetic field may contribute substantially to the noise. Subsequently, Sisodia and Gandhi (1966), Sisodia, Gandhi, and Wadhwa (1967) and Sisodia and Wadhwa (1968) found that large reductions in the noise output of a crossed-field gun could be achieved by either tilting the cathode slightly in the magnetic field direction or by slightly perturbing the magnetic field in the gun region to give small magnetic field components perpendicular to the gun. The noise reduction is believed to be due to smoothing out of the random velocities in the magnetic field direction, although this conclusion must be considered tentative until further three-dimensional theoretical and/or experimental work is completed.

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APPENDIX B

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COMPUTER AIDED DESIGN AND ANALYSIS OF ELECTRON GUNS FOR INJECTED BEAM CROSSED FIELD AMPLIFIERS

Elden K. Shaw and Gerald P. Kooyers

Presented at the 1977 International Electron Devices Meeting, Washington, D.C. ant ulud ou thisquerit filles of there a so a second a g

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COMPUTER AIDED DESIGN AND ANALYSIS OF ELECTRON GUNS FOR INJECTED BEAM CROSSED FIELD AMPLIFIERS

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Palo Alto, California

ABSTRACT

In this paper a self-consistent computer aided CTA gum analysis program is discussed which can be used to solve for the characteristics of "long" crossed field guns with arbitrary electrode shapes and including the 235 effects of space charge. The results of analyzing a "long" Kino gun are presented which show excellent agreement between the theoretical gun parameters and those obtained by the computer analysis. As an example of the design procedure, the results of designing a CFA gun and transition region with the accelerator at mode potential are given. The utility of the model is further illustrated by the results of some preliminary noise calculations made by simulating the noise characteristics of the beam at the cathode surface.

INTRODUCTION

In order to build efficient, high power injected bean crossed field amplifiers it is necessary to be able to design convergent crossed field guns which produce relatively thin laginar beams. The design and analysis of guns for practical CFA's is complicated due to the cycloidal motion of the electrons which leads to trajectory crossings and thus very complicated flow.

In the past CTA guns have been designed using empirical techniques or the design methods suggested by Kino(1) with potential plotting used to match equipotential lines in the transition region between the gun and the interaction region.

A self-consistent CTA gun analysis procedure is discussed below which can be used to solve for the characteristics of "lorg" crossed field guns with arbitrary electrode shapes. The effects of space charge are included in the analysis. In addition, the analysis procedure discussed below shows promise of being able to be used to study noise unchanisms and noise growth in the gun and transition region.

TWO DIMENSIONAL LAGRANGIAN MODEL

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In this approach a large number (up to 6000) rods of charge are used to simulate the beam in the gun and transition regions with rods being released from the cathode consistent with the emission characteristics of the cathode surface.

The approach is discussed below:

(1) First, the gun and transition region is and divided into a rectangular mesh.

- (2) Next, the potentials in the absence of space charge are calculated at each mesh point using a computer code which solves LaPlace's equation for arbitrary electrode shapes and potentials. These potentials are then stored for use during the rest of the calculation.
- (3) A relatively large number of rods of charge are released from the cathode with position, density, and velocity consistent with the emission characteristics of the cathode. The beam buildup is followed in time with electrons being emitted from the cathode at equal small increments in time. The beam is allowed to build up to several thousand rods of charge with "new" Toda of charge added at the cathode in each time step and "old" rods being removed as they exit the right-hand boundary or are collected on the electrodes.
- (4) The motion of the rods of charge is governed by the Lorentz Force Equation where the electric field is found from the potentials on the mesh points. During each iteration, each rod of charge is allowed to move for a small increment in time with its motion governed by the Lorentz Force Equation. Very accurate finite difference equations are used which have previously been developed and used successfully to solve the Lorents Force Equation(2).

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- (5) Simulating the beam by a large number of rods of charge has not been used in the past as an approach to the design of crossed field guns due to the relatively long computer times required to solve Poisson's Equation by conventional techniques. However, a technique due to R. Mockney(3) has been successfully used to solve Poisson's Equation in less than one second on an IBM 360/168 (compared with approximately one minute for the usual iterative relaxation procedures). This approach is discussed in some detail in a paper by Yu, Kooyers and Buneman(2) and was used successfully with up to four thousand rods of charge in analyzing for the first time the distributed emission CFA. Once the potentials have been calculated including space charge by the Hockney technique for each iteration, "new" rods of charge are released from the cathode and the procedure discussed above is repeated until the beam reaches a "steady state." Steady state or convergence is obtained when the number of rods of charge exiting the gun region averaged over one cyclotron period approaches a constant.

GUN STUDIES USING THE MODEL

Long Kino Gun Analysis

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In order to test the accuracy and validity of the model a long Kino gun was designed with the following parameters:

Magnetic Field - .250 w/m² Cathods Length - .150 in. Normalized Length - 116.4 Kino units Accelerator Potential - 5000 volts Cathods Current Density - 14 amp/cm² Total Beam Current - 3.34 amo.

Using a time step of 1/20 of a cyclotron period, the beam buildup characteristics were calculated for 110 iterations. As rods of charge were injected at the cathode surface the beam current as measured by the rods of charge exiting the gun region rapidly built up to an average value of 3,24 maps which differs from the Kino predicted value by less than 3%. The beam thickness and current density of the exiting beam are also in close agreement with the theoretical values given by the Kino theory.

Gun Design with Accelerator at Cathode Potential

The computer model becomes most useful as a design tool for guns when used to design both the gun and transition regions. For a particular dual-mode CFA a gun was required to operate at a magnetic field of $.3 \text{ v/m}^2$ and produce a current of 5 amps.

As a starting point, a Kino gun design was made with the following characteristics:

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Current Density - 14 amps/cm² Cathode Length - .150 inches Kino Calculated Current - 5.7 amps Anode Potential - 8.2 kv.

Running the accelerator at full anode potential represents an innovation made possible by recognizing that in a gridded gun (this design war made for the eventual inclusion of a grid), grid potential can be used to vary the catho current. Thus, a separate accelerator is not necessary. Eliminating the accelerator resu three immediate advantages:

- (1) Reduction of the number of required electrode voltages by one.
- (2) Elimination of the gap between the accelerator and the snode, thus reducing the length required for the transition region which in turn reduces diocotron growth and gun noise.
- (3) Elimination of the potential and field matching problems due to the accelerator and anode gap.

Although the cathode current fluctuates, the average value given by the computer model is about 5.0 amps which is only slightly less than that predicted by the Kino theory. However, it must be kept in mind that the accelerator electrode is not the exact Kino shape due to the necessary compromises made to match up the snode and the accelerator and thus it is not expected that the exact Kino current would be obtained.

The beam thickness varies with position, but an average value of about 10 mils is obtained for the exiting beam. This design was reached only after several sole, beam focusing electrode, and mode shapes were analyzed. Only with a computer model which is capable of modeling the complete gun region back to the cathode, can designs such as this be made with any confidence.

Moise Simulation in the Crossed Field Beam

In order to study the noise properties of the beam, it is necessary to simulate the emission properties of the hot cathode surface as accurately as possible. For the computer model this means that the following parameters must be determined st the cathode surface:

- (1) The number of electrons emitted.
- (2) The emission velocities of the electrons.
- (3) The time of emission of the electrons.

The method for calculating each of the above three quantities is summarized below:

(1) Inission Number

It can be shown that the probability of exactly S electrons being emitted from a given cathode spot of length Δy during the time interval Δt is given by

$$f(S) = \frac{e^{-\lambda_{\lambda}S}}{S!}.$$

where λ is the average number of electrons emitted in each Δt . This is recognized to be the Poisson Distribution.

The probability that 5 or fewer electrons are emitted in the time interval at is given by the cumulative distribution function

$$\mathbf{Y}(S) = \sum_{S=0}^{S} f(S)$$

and

To determine the number of electrons emitted in a time interval, Δt , a uniformly distributed random number between 0 and 1 is generated. The number of electrons emitted is 5 if

F(S-1) < 1 4 F(S)

When $R \leq F(0)$ no electrons are emitted.

(2) Emission Velocities

A half-Mamwellian distribution is assumed for the x component of emission velocity (the component perpendicular to the cathode). The x velocity is then given by

$$\dot{x} = \left(\frac{2kT}{n}\right)^{l_{\chi}} \left(-\ln R_{\chi}\right)^{l_{\chi}}$$

where R_x is a random number uniformly distributed between 0 and 1, T is the cathode temperature, k is Boltzmann's constant, and m is the electron mass.

A full-Maxwellian distribution is assumed for electron velocities in the y direction (parallel to the cathode). The relation giving this velocity is

$$\dot{y} = \left(\frac{2kT}{n}\right)^{h} \operatorname{erf}^{-1}(R_{y})$$

where R is a random number uniformly distributed between 0 and 1.

(3) Emission Times

In order to avoid calculating the positions of the electrons above the cathode for random emission times, it is assumed that all S of the electrons in a given mesh region are emitted at the same time but with a random distribution over the area. The emission points are then given by

$$y = y_0 \pm R_y \frac{\Delta y}{2}$$

where R is a random number between 0 and 1 and the sign of R is also selected randomly. Thus, to an observer looking at one point on the cathode surface the electrons will appear to come off that spot at random times. Provided the time interval Δt is chosen to be relatively small, the degree of approximation introduced should be small.

Preliminary calculations of beam "noisiness" as measured by velocity variance and noise power spectral density calculations have led to the following preliminary observations:

- (1) Operation of a CTA gun in the temperature limited regime leads to a marked increase in beam noise. Observation of the beam shape and trajectories during temperature limited operation shows a deterioration of the gun optics which may explain the increased noise.
- (2) The noise grows approximately exponentially with distance in the direction of beam flow. This result is not unexpected since amplification of the noise due to the diocotron effect would be expected to be approximately exponential.
- (3) The upper edge of the beam (nearest the anode) is noisier than the lower edge of the beam.

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APPENDIX C

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COMPUTER-AIDED DESIGN OF ELECTRON GUNS FOR INJECTED-BEAM CROSSED-FIELD AMPLIFIERS

Elden K. Shaw and Gerald P. Kooyers

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Computer-Aided Design of Electron Guns for Injected-Beam Crossed-Field Amplifiers

ELDEN K. SHAW AND G. P. KOOYERS

Abstruct-A two-dimensional self-consistent computer-aided CFA gun analysis program is discussed which can be used to solve for the characteristics of "long" crossed-field guns with arbitrary electrode shapes and including the effects of space charge. The results of analyzing a "long" Kino gun show excellent agreement between the theoretical gun parameters and those obtained by the computer analysis. The model shows promise of also being useful in the analysis of noise in CFA guns.

I. TWO-DIMENSIONAL LAGRANGIAN MODEL

In this approach a large number (up to 6000) of rods of charge are used to simulate the beam in the gun and transition regions with rods being released from the cathode consistent with the emission characteristics of the cathode surface.

The approach is summarized below:

1) First, the gun and transition region is divided into a rectangular mesh.

2) Next, the potentials in the absence of space charge are calculated at each mesh point using a computer code which solves Laplace's equation for arbitrary electrode shapes and potentials. These potentials are then stored for use during the rest of the calculation.

3) A number of rods of charge are released from the cathode with position, density, and velocity consistent with the emission characteristics of the cathode. The beam buildup is followed in time with electrons being emitted from the

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The authors are with UCA Systems, Inc., Palo Alto, CA 94304.

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Fig. 3. Beam density profile for the "long" Kino gun. Notes: 1) There are 2555 rods of charge simulating the beam. 2) 152 rods of charge are injected at each time step; only an average of 21 exit the gun region. 3) The current exiting to the right differs from the Kino calculated value by only 5 percent. 4) Random electron thermal velocities are included in this calculation. 5) Electron shapes are from Kino gun theory.

cathode at equal small increments in time. The beam is allowed to build up to several thousand rods of charge with "new" rods of charge added at the cathode in each time step the potentials on the mesh points. During each iteration, and "old" rods being removed as they exit the right-hand each rod of charge is allowed to move for a small increment in boundary or are collected on the electrodes.

4) The motion of the rods of charge is governed by the Lorentz force equation where the electric field is found from time with its motion governed by the Lorentz force equation.

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IEEE TRANSACTIONS ON ELECTRON DEVICES, VOL. ED-26, NO. 7, JULY 1979

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5) Simulating the beam by a large number of rods of charge has not been used in the past as an approach to the design of crossed-field guns due to the relatively long computer times required to solve Poisson's equation by conventional techniques. However, a technique due to Hockney [1]-[3] has been successfully used to solve Poisson's equation in less than 1 s on an IBM 370/168 (compared with approximately 1 min for the usual iterative relaxation procedures). This approach is discussed in some detail in a paper by Yu, Kooyers, and Buneman [4] and was used successfully in analyzing for the first time the distributed emission CFA. Once the potentials have been calculated including space charge by the Hockney technique for each iteration, "new" rods of charge are released from the cathode and the procedure discussed above is repeated until the beam reaches a "steady state." Steady state or convergence is obtained when the number of rods exiting the gun region averaged over one cyclotron period approaches a constant.

II. ANALYSIS OF A LONG KINO GUN

In order to determine the validity of the computer model, the "long" Kino gun [5] shown in Fig. 1 was analyzed. The parameters of this gun are:

total beam current	3.34 A
cathode length	0.150 in
magnetic field	0.2500 W/m ²
current density	i4 A/cm ²
accelerator potential	5000 V
normalized cathode length	116.4 Kino units.

Current exiting the gun region as a function of time step is shown in Fig. 2. For this case the current rises rapidly and then fluctuates about a mean value which is close to the Kino predicted value. In this case, the cathode is operating spacecharge-limited with only a fraction of the total emitted rods exiting the gun region. Shown in Fig. 3 is the exact Kino beam shape compared to that obtained by plotting the individual charge positions from the computer program. Again, the agreement between the computer-calculated results and the Kino theory is good. In Fig. 4 the computer-calculated using the Kino equations.

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APPENDIX D

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SOLVING POISSON'S EQUATION IN THREE DIMENSIONS D.1 Solving Poisson's Equation in Three Dimensions

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We want to solve Poisson's Equation in the region sketched in Figure D-1 with a constant potential, ϕ_{C} , on the z boundaries.

First, the potential, ϕ_C , is subtracted from all boundary potentials since this results in simpler boundary conditions (i.e., two boundaries at zero potential rather than one). The modified boundary conditions are shown in Figure D-1.

A coordinate grid system is introduced with 1 segments in the x-direction, m in the y-direction, and n in the z-direction as sketched in Figure D-1.

For the mesh system defined above the boundary conditions are now given by

 $\phi_{i,j,1} = \phi_{i,j,n+1} = 0$ $\phi_{1,j,k} = -\phi_C$ $\phi_{\ell+1,j,k} = \phi_A - \phi_C = \phi_B$ $\phi_{1,0,k} = \phi_{1,2,k}$ $\phi_{i,m,k} = \phi_{i,m+2,k}$ $\frac{\partial \phi}{\partial n} = 0$ (D-1)

The five point difference approximation to Poisson's Equation is:

 $\frac{1}{\Delta x^{2}} \quad \phi_{i-1,j,k} = 2\phi_{i,j,k} + \phi_{i+1,j,k} + \frac{1}{\Delta y^{2}} \quad \phi_{i,j-1,k} = 2\phi_{i,j,k} + \phi_{i,j+1,k} + \frac{1}{\Delta z^{2}} \quad \phi_{i,j,k-1} = 2\phi_{i,j,k} + \phi_{i,j,k+1} = \frac{\rho_{i,j,k}}{\epsilon_{0}} \qquad (D-2)$



Figure D-1. Three-Dimensional Geometry with Modified Boundary Conditions for the Case of Constant Potential on z-Boundary.



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Figure D-2. Boundary Conditions for the Two-Dimensional Helmholtz Equation.

where the subscripts i, j, and k refer to the point (x_i, y_j, z_k) such that

$$x_{i} = (i-1)\Delta x$$
$$y_{j} = (j-1)\Delta y$$
$$z_{k} = (k-1)\Delta z$$

and

$$\Delta y = \frac{b}{m}$$
$$\Delta z = \frac{c}{n}$$

 $\Delta \mathbf{x} = \frac{\mathbf{a}}{\mathbf{l}}$

We now take the finite Fourier expansion of $\phi_{i,j,k}$ in the k (i.e., z) direction with the boundary conditions

$$\phi_{i,j,1} = \phi_{i,j,n+1} = 0$$

For $l \leq k \leq n-1$ we have

$$\overline{\phi}_{i,j,k+1} = \sum_{s=1}^{n-1} \phi_{i,j,s+1} \sin \frac{\pi s_k}{n}$$
 (D-4)

(D-3)

where $\overline{\phi}$ is the Fourier transform of ϕ and

$$\phi_{i,j,k} = \frac{2}{n} \sum_{s=1}^{n-1} \frac{1}{\phi_{i,j,s+1}} \sin \frac{\pi s (k-1)}{n}$$
 (D-5)

k = 2, 3, ..., n.

Substituting the expression for ϕ into the three-dimensional five point difference equation, simplifying and equating the sine components, we obtain the equation

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$$\overline{\overline{\rho}}_{i,j,s+1} = \frac{1}{\Delta x^2} \overline{\phi}_{i-1,j,s+1} - 2\overline{\phi}_{i,j,s+1} + \overline{\phi}_{i+1,j,s+1}$$

$$+ \frac{1}{\Delta y^2} \overline{\phi}_{i,j-1,s+1} - 2\overline{\phi}_{i,j,s+1} + \overline{\phi}_{i,j+1,s+1} + \frac{2}{\Delta z^2} \cos \frac{\pi s}{n} - 1$$

$$\overline{\phi}_{i,j,s+1} \qquad (D-6)$$

where we have also used the sine expansion for ρ . We note that this is just the finite difference approximation to the Helmholtz equation in <u>two</u> dimensions for each value of s. Thus, in order to solve the three-dimensional Poisson Equation, we use the following procedure:

(1) Calculate \$\vec{\rho}\$_{i,j,k}\$ for all i,j by analyzing \$\rho\$_{i,j,k}\$ along k.

$$\frac{1}{\rho} \frac{n-1}{1, j, k+1} = \sum_{a=1}^{n-1} \frac{1}{1, j, s+1} \sin \frac{\pi_s(k-1)}{n}$$
 (D-7)

with the $\rho_{i,j,k}$'s being known.

This will require applying a Fourier Analysis program (t-1)x(m-1)x(n-1) times. To accomplish this analysis, an efficient computer program called FOUR67 is used.

(2) Knowing the $\overline{\rho}$'s we then solve the two-dimensional Helmholtz equation

$$\nabla^2 \overline{\phi} + K^2 \overline{\phi} = \overline{\rho}$$

for each of the <u>n-1</u> values of K. This two-dimensional problem is solved using a computer program called PWSCRT.

In order to solve the Helmholtz equation the boundary conditions of the transformed potentials must be known. We note that since

$$\phi_{i,j,k} = -\phi_C$$

. .

and $\phi_{i+1,j,k} = \phi_A^{-\phi} C$

then
$$\overline{\phi}_{1,j,k} = -\phi_C \sum_{s=1}^{n-1} \sin \frac{\pi s (k-1)}{n}$$
 (D-8)

and
$$\overline{\phi}_{l+1,j,k} = (\phi_A - \phi_C) \sum_{s=1}^{n-1} \sin \frac{\pi s (k-1)}{n}$$
 (D-9)

The boundary conditions at j=l and j=m+l are

$$\frac{\partial \phi}{\partial y} = 0$$
or $\phi_{1,0,k} = \phi_{1,2,k}$
and $\phi_{1,m,k} = \phi_{1,m+1,k}$

which give

$$\frac{\partial \overline{\phi}_{1,\underline{w}},\underline{h}_{k}}{\partial y} = \frac{\partial \overline{\phi}_{1,\underline{1},\underline{k}}}{\partial y} = 0 \qquad (D-10)$$

These boundary conditions are sketched in Figure D-2 for the k-th row in the z-direction.

(3) Given the \$\overline\$, j,k\$ we then use the inverse Fourier Transform (Synthesis)

$$\phi_{i,j,s+1} = \frac{2}{n} \sum_{k=1}^{n-1} \phi_{i,j,k+1} \sin \frac{(\pi s_k)}{n}$$
 (D-11)

to obtain the potentials on all of the interior points. The efficient computer program FOUR67 is used to perform the above synthesis on the (l-1)x(m-1)x(n-1) potential points. APPENDIX E

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FINITE DIFFERENCE EQUATIONS OF MOTION IN THREE DIMENSIONS

E.1 EQUATIONS OF MOTION

In order to find the motion of the charges, it is necessary to solve the equation

$$m \frac{d\vec{v}}{dt} = q(\vec{E} + \vec{v} \times \vec{B}) .$$

Assuming both electric and magnetic field components in the x, y, and z directions, the components of the above equation become

 $\ddot{\mathbf{x}} = \frac{\mathbf{q}}{\mathbf{m}} \left(\mathbf{E}_{\mathbf{x}} + \dot{\mathbf{y}} \mathbf{B}_{\mathbf{z}} - \dot{\mathbf{z}} \mathbf{B}_{\mathbf{y}} \right)$ (E-1)

$$\ddot{\mathbf{y}} = \frac{\mathbf{q}}{\mathbf{m}} \left(\mathbf{E}_{\mathbf{y}} + \dot{\mathbf{z}} \mathbf{B}_{\mathbf{x}} - \dot{\mathbf{x}} \mathbf{B}_{\mathbf{z}} \right)$$
(E-2)

$$\dot{z} = \frac{q}{m} \left(E_z - \dot{y} B_z + \dot{x} B_y \right)$$
(E-3)

-Assuming that the electric and magnetic fields are constant in the interval of integration, the sove equations can be integrated once to obtain:

$$\dot{x}(t) - \dot{x}(0) = \frac{q}{m} \left[E_{x} t + (y(t) - y(0))B_{z} - (z(t) - z(0))B_{y} \right]$$
(E-4)

$$\dot{y}(t) - \dot{y}(0) = \frac{q}{m} \left[E_y t + (z(t) - z(0))B_x - (x(t) - x(0))B_z \right]$$
(E-5)

$$\dot{z}(t) - \dot{z}(0) = \frac{q}{m} \left[E_z t - (y(t) - y(0))B_z + (x(t) - x(0))B_y \right]$$
(E-6)

Using the Laplace Transform on Eqs. E-4, E-5 and E-6, we obtain the linear equations

$$\tilde{\mathbf{x}} - \omega_{cz}\tilde{\mathbf{y}} + \omega_{cy}\tilde{\mathbf{z}} = n \frac{E_{x}}{s^{2}} + \frac{1}{s} (\omega_{cy}z_{0} - \omega_{cz}y_{0} + \dot{\mathbf{x}}_{0}) + \mathbf{x}_{0}$$
(E-7)

$$\omega_{cz}\tilde{x} + s\tilde{y} - \omega_{cx}\tilde{z} = n \frac{E_{y}}{s^{2}} + \frac{1}{s} (\omega_{cz}x_{0} - \omega_{cx}z_{0} + \dot{y}_{0}) + y_{0}$$
(E-8)
$$- \omega_{cy}\tilde{x} + \omega_{cx}\tilde{y} + s\tilde{z} = n \frac{E_{z}}{s^{2}} + \frac{1}{s} (\omega_{cx}y_{0} - \omega_{cy}x_{0} + \dot{z}_{0}) + z_{0}$$
(E-9)

where
$$\omega_{cx} = nB_{x}$$

 $\omega_{cy} = nB_{y}$
 $\omega_{cz} = nB_{z}$
 $n = \frac{q}{m}$
 $x_{0} = x(0)$ $\dot{x}_{0} = \dot{x}(0)$
 $y_{0} = y(0)$ $\dot{y}_{0} = \dot{y}(0)$
 $z_{0} = z(0)$ $\dot{z}_{0} = \dot{z}(0)$

and \tilde{x} , \tilde{y} , \tilde{z} are the Laplace Transforms of x(t), y(t), z(t).

Equations E-7, E-8 and E-9 can be solved simultaneously for \tilde{x} , \tilde{y} , \tilde{z} and then inverse transformed to obtain

$$\mathbf{x} - \mathbf{x}_0 = \frac{\dot{\mathbf{x}}_0}{\omega_c} \left[\sin \tau + \frac{\omega_{cx}^2}{\omega_c^2} \left(\tau - \sin \tau \right) \right] +$$

$$\frac{\dot{\mathbf{y}}_{0}}{\omega_{c}} \begin{bmatrix} \frac{\omega_{cz}}{\omega_{c}} \left(1 - \cos\tau\right) + \frac{\omega_{cx}}{\omega_{c}} & \frac{\omega_{cy}}{\omega_{c}} \left(\tau - \sin\tau\right) \end{bmatrix} \\ + \frac{\dot{z}_{0}}{\omega_{c}} \begin{bmatrix} -\omega_{cy}}{\omega_{c}} \left(1 - \cos\tau\right) + \frac{\omega_{cx}}{\omega_{c}} & \frac{\omega_{cz}}{\omega_{c}} \left(\tau - \sin\tau\right) \end{bmatrix}$$





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MICROCOPY RESOLUTION TEST CHART NATIONAL BUREAU OF STANDARDS-1963-A :

$$+ \frac{nE_{x}}{\omega_{c}^{2}} \left[\frac{\omega_{cx}}{\omega_{c}^{2}} \left(-1 + \frac{\tau^{2}}{2} + \cos \tau \right) \right]$$

$$+ \frac{nE_{y}}{\omega_{c}^{2}} \left[\frac{\omega_{cx}}{\omega_{c}} \left(\frac{\omega_{cy}}{\omega_{c}} \left(-1 + \frac{\tau^{2}}{2} + \cos \tau \right) + \frac{\omega_{cz}}{\omega_{c}} \left(\tau - \sin \tau \right) \right]$$

$$+ \frac{nE_{z}}{\omega_{c}^{2}} \left[\frac{\omega_{cx}}{\omega_{c}} \left(\frac{\omega_{cz}}{\omega_{c}} \left(-1 + \frac{\tau^{2}}{2} + \cos \tau \right) - \frac{\omega_{cy}}{\omega_{c}} \left(\tau - \sin \tau \right) \right]$$

$$(E-10)$$
where $\omega_{c}^{2} = \omega_{cx}^{2} + \omega_{cy}^{2} + \omega_{cz}^{2}$

Equations for y and z are not given since they are easily obtained from Equation E-10 by permuting the coordinates as given below:

> x + y y + z z + x

and $\tau = \omega_{ct}$.

In the finite difference procedures used in the computer program, the x, y, and z positions of the particles are found at $t + \Delta t$ using Equation E-7 and the similar equations for y and z. Equations E-4, E-5, and E-6 are then used to find the x, y, and z velocities at $t + \Delta t$.



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SAMPLE COMPUTER CALCULATED OUTPUT FROM THE THREE-DIMENSIONAL PROGRAM

100.0000-TH PARTICLE 20.0000-TH TIME STEP FOR EVERY TRAJECTORY INFORMATION WILL BE PRINTED OUT EVERY

----PRINT PARAMETERS----

20.000

20.000 20.0000 20.0000 20.0000 20.000 MAXIMUM NUMBER OF EXITING TRAJECTORIES SAVED FOR EACH BOUNDARY

CHARGE NORMALIZATION CONSTANT = 0.1270E+10

THE Z-BOUNDARY POTENTIAL = 0.0 POTENTIAL NORMALIZATION CONSTANT = 0.1409E+00

.35386+04

----PARTICLE EXIT PARAMETERS----

.3391E+M 0.3410E+M 0.3443E+M 0.3466E+M 0.3403E+M 0.3495E+M 0.3502E+04 0.3506E+04 0.3510E+04 0.3515E+04 0.3520E+04

-2263E+M 0.2322E+M 0.2300E+M 0.2439E+M 0.2490E+M 0.259E+M 0.2617E+M 0.2676E+M 0.273E+M 0.2786E+M 0.2843E+M 0.2845E+M

THE MARER OF PARTICLES IN THE BRILLOUIN BEAM IS = 0.3000E405 THE SPACE CHARGE CALCULATION PARAMETER IS 0.0 Array of Boundary Potentials at X=A 0.9050E403 0.1154E404 0.1364E404 0.1495E404 0.153E404 0.1053E404 0.1938E404 0.1996E404 0.2006F404 0.2201E404

----POTENTIAL AND SPACE CHARGE PARAMETERS-

1.000

RANDON VELOCITY PARAMETER =

18.8009

IN Z-DIRECTION

42.0004 17.0000

= 10.0000 NUMBER OF CATHODE SEGMENTS IN Z-DIRECTION 323.0000 NUMBER OF PARTICLES EMITTED PER MESH REGION = Random Postion Parameter = 1.0000 RANDOM VELOCITY P

1323.000

NUMBER OF CATHODE SEGNENTS IN Y-DIRECTION = Cathode Temperature in Degrees Kelvin = 132; Randon Number Parameter = 1.0000 Rah

5.000 3.000

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ENDING Y-HESH POINT OF CATHODE = ENDING Z-HESH POINT OF CATHODE =

-CATHODE ENISSION PARAMETERS----

THE MAXIMUM MUMBER OF PARTICLES = 19000.0000

0.2500

THE Z-DIRECTED NAMETIC FIELD =

10.000

THE X-DIRECTED MARKITC FIELD = 0.0 THE MARDER OF TIME STEPS PER CYCLOTRON PERIOD OUTPUT TRAJECTORY NORMALIZATION PARAMETER = THIS CALCULATION PROCEEDS FOR 76.0000 ITERA

76.0000 ITERATIONS

----TRAJECTORY PARAMETERS-

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36.000 14.00

M DIVISIONS IN THE X-DIRECTION = 32. M DIVISIONS IN THE Y-DIRECTION = 40. M DIVISIONS IN THE Z-DIRECTION = 16. THE GLN IN THE X-DIRECTION IN INCHES = THE GLN IN THE Y-DIRECTION IN INCHES =

ND. OF NESH DIVISIONS IN THE X-DIRECTION = 36 ND. OF NESH DIVISIONS IN THE Y-DIRECTION = 40 ND. OF NESH DIVISIONS IN THE Z-DIRECTION = 16 NEIGHT OF THE GLN IN THE X-DIRECTION IN INCHES = LENGTH OF THE GLN IN THE Y-DIRECTION IN INCHES = NIDTH OF THE GLN IN THE Z-DIRECTION IN INCHES =

4.0000

HHHHH TEST CASE FOR GUNDO 7/5/79

----GEONETRY PARAMETERS----

-2%7E+04 0.2997E+04 0.3045E+04 0.3090E+04 0.3132E+04 0.3172E+04 0.3209E+04 0.3243E+04 0.3275E+04 0.3395E+04 0.3334E+04 0.3363E+04

25.0000-TH TINE STEP : : : r -1.0000-TH TIME STE 100.0000-TH TIME STEP 1.0000 THE SPACE CHARGE MATRIX MILL BE PRIMIED GUT EVERY -1.0000-TH TIME STE THE POTENTIAL MATRIX MILL BE PRIMIED GUT EVERY 100.0000-TH TIME STEP TRAJECTORY INFORMATION MILL BE PRIMIED GUT FOR EXITING PARTICLES EVERY THE PARAMETER HAICH CONTROLS EXITING PARTICLE PRIMI GUT IS = 1.0000 BEAM PROFILE PLOTS MILL BE MUDE EVERY 15.0000-TH TIME STEP MARY OF Z-VALUES FOR PROFILE PLOTS ARAY OF VIDTH OF PROFILE PLOTS ARAY OF WIDTH OF PROFILE PLOTS ARAY OF VIDTH OF PROFILE PLOTS ARAY OF VALUES FOR PROFILE PLOTS ARAY OF VALUES FOR PROFILE PLOTS ARAY OF VALUES FOR PROFILE PLOTS ARAY OF Y-VALUES FOR PROFILE PLOTS ARAY OF Y-VALUES FOR PROFILE PLOTS

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Ĩ	9.13066+02	-0.14036+02	-0.14106+02	-0.14336+02	-0.14476+02	-0.1460E+02 -	-0.1472E+02	-6.1484E+02 -	-0.1494E+02 -	-0.1504E+02	
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BEAN PROFILE PLOT FOR ITERATION ND. 75 AT THE X-NEW POINT 16 NITH X-THICKNESS OF

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75 IS 15494 THE MAXIMUM NUMBER OF PARTICLES USED = 15674 THE TOTAL NUMBER OF PARTICLES FOR ITERATION NUMBER

TN = 3.4162E+02 ZN = 0.1342E+02 ZN = 0.1274E+02 0.12496+02 PARTICLE NO. 136 WITH AGE 1 EXITS THE LOWER BOUNDARY MITH XN = 0.2492E+01 YN = 0.4225E+02 ZN = 0.0656E+01 VXN = -0.1225E-01 VXN = -0.2054E-01 VZN = -0.1223E-01 0.1268E+02 0.12436+02 YN = 0.4174E+02 ZN = 0.6676E+01 0.8468E+01 ZN = 0.4829E+01 = 7 -N TN = 0.42026402 ZN = R YN = 0.4167E+02 PARTICLE ND. 12 NITH AGE 1 EXITS THE LONER BOUNDARY NITH XM = 0.2064E401 YN = 0.4154E402 VXM = -0.1722E401 VYM = -0.9301E-01 VZM = -0.2367E-02 0.41766+02 TN = 0.41946+02 TN = 0.4170E+02 # K ARTICLE ND. 13 MITH AGE 1 EXITS THE LONER BOUNDARY NITH XM = 0.2244E+01 VXM = -0.9628E+00 VYM = -0.1002E-01 VZM = -0.1272E-02 MITH AGE 1 EXITS THE LOWER BOUNDARY NITH XN = 0.2346E+01 VYN = -0.2968E-01 VZN = 0.8326E-02 PARTICLE ND.4 16 WITH AGE 1 EXITS THE LONER BOUNDARY WITH XN = 0.2453E+01 VXN = -0.2996E+00 VYN = -0.6106E-01 VZN = -0.5252E-02 MITH AGE 1 EXITS THE LOWER BOUNDARY WITH XN = 0.2195E+01 VYN = -0.7884E-01 VZN = 0.1171E-01 PARTICLE NO. 160 MITH AGE 1 EXITS THE LOWER BOUNDARY MITH XM = 0.2335E+01 VXM = -0.6647E+00 VYM = -0.1356E-01 VZM = -0.3844E-02 PARTICLE NO. 175 NITH AGE 1 EXITS THE LOWER BOUNDARY WITH XN = 0.1462E+01 VXN = -0.3212E+01 VYN = -0.2330E+00 VZN = 0.5171E-02 \ARTICLE ND. 14 MITH AGE 1 EXITS THE LOWER BOUNDARY MITH XN = 0.2037E+01 VXN = -0.1560E+01 VTN = -0.1010E+00 VZN = 0.9057E-02 ARTICLE ND. 135 | VXN = -0.5984E+00 VXN = -0.1040E+01 ñ PARTICLE ND. PARTICLE NO. PARTICLE NO. PARTICLE ND.

PARTICLE ND. 182 MITH AGE 1 EXITS THE LONER BOUNDARY MITH XM = 0.1660E+01 YM = 0.4179E+02 ZM = 0.4631E+01 VXM = -0.2809E+01 VYM = -0.2459E+00 VZM = 0.1150E-01

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PARTICLE HD. 530 MITH AGE 11 EXITS THE RIGHT BOUNDARY MITH XN = 0.1350E+02 YN = 0.5080E+02 XN = 0.5255E+01 VXN = 0.3113E+00 VYN = 0.3249E+01 VZN = 0.1314E-01 PARTICLE ND. 106 NITH AGE 1 EXITS THE LONGR BOUNDARY WITH XM = 0.2286E401 YM = 0.4211E402 ZM = 0.3112E401 YXM = -0.0347E400 YYM = -0.9145E-01 YZM = 0.1552E-01 0.4462E+01 0.6010E+01 0.1746E+02 0.1570E+02 0.1536E+02 0.1203E+02 0.1523E+02 0.1391E+02 0.1327E+02 0.4060E+01 0.1290E+02 0.4023E+01 0.7827E+01 0.3972E+01 0.5972E+01 0.8045E+01 0.9367E+01 0.4920E+0 # 7 8.5044E+02 ZN = -R = N . -R * 72 -7 -" R # R 0.4072E+02 ZN = -R z R MATICLE ND. 270 MITH AGE 7 EXITS THE LONER BOUNDARY MITH XH = 0.2362E401 YH = 0.4247E402 VXH = -0.1050E401 VYH = 0.2435E400 VZH = 0.1769E-01 PARTICLE ND. 357 NITH AGE 11 EXITS THE RIGHT BOUNDARY NITH XM = 0.1425E+02 YM = 0.5123E+02 VXM = 0.3652E+00 VYM = 0.3594E+01 VZM = 0.2451E-01 0.5095E+02 0.4090E+02 TN = 0.4074E+02 MITH AGE 11 EXITS THE RIGHT BOUNDARY MITH XN = 0.1501E+02 YN = 0.5139E+02 VYN = 0.4261E+01 VZN = -0.3245E-01 TN = 0.4173E+02 TN = 0.4100E+02 0.5044E+02 0.5027E+02 TN = 0.4220E+02 YN = 0.4112E+02 0.51952+02 0.5121E+02 0.5078E+02 9.41486+02 ARTICLE ND. 299 NITH AGE 1 EXITS THE LAWER BOUNDARY NITH XN = 0.2368E401 YN = VXN = -0.4404E400 VYN = -0.1301E-01 VZN = -0.1313E-01 ARTICLE NO. 200 MITH AGE 13 EXITS THE RIGHT BOUNDARY MITH XH = 0.1136E+02 YH = VXH = -0.7371E+00 VYH = 0.3296E+01 VZH = -0.5945E-02 **.** 7 # K N = 0.1402E+02 YN = ARTICLE NO. 496 MITH AGE 8 EXITS THE RIGHT BOUNDARY MITH XN = 0.1170E+02 YN = VXN = -0.0372E+00 YYN = 0.2942E+01 VZN = 0.1786E+00 ARTICLE ND. 300 NITH AGE 1 EXITS THE LONER BOLNDARY NITH XN = 0.2495E+01 YN = VXN = -0.3494E-01 VTN = -0.1929E-01 VZN = 0.7343E-02 * Z * ¥ RIGHT BOUNDARY NITH XN = 0.1211E+02 0.7419E-01 WITH AGE 1 EXITS THE LOWER BOUNDARY WITH XM = 0.1730E+01 VTN = -0.1782E+00 VZN = -0.3778E-02 MRTICLE NO. 441 MITH AGE 5 EXITS THE LONER BOUNDARY NITH XN = 0.19782401 VXN = -0.25172401 VYN = -0.00012-01 VZN = -0.56672-02 PARTICLE ND. 516. NITH AGE 8 EXITS THE RIGHT BOUNDARY NITH XN = 0.1526E+02 VXM = 0.2077E+01 VYM = 0.3747E+01 VZN = 0.6763E-01 MATICLE ND. 187 WITH AGE 1 EXITS THE LOWER BOUNDARY WITH XN = 0.18356401 VAN = -0.22026401 VTH = -0.16346400 VZH = -0.73956-02 ARTICLE NO. 92 MITH AGE 11 EXITS THE RIGHT BOUNDARY NITH XN = 0.1608E402 VXN = -0.1578E400 VYN = 0.4260E401 VZN = -0.2981E-01 ARTICLE ND. 501 MITH AGE 13 EXITS THE RIGHT BOUNDARY NITH XN = 0.1686E402 VXN = 0.1688E400 VYN = 0.4278E401 VZN = 0.3601E-02 WITH AGE 3 EXITS THE LOWER BOUNDARY WITH XM = 0.2227E+01 VTM = -0.1019E+00 VZN = 0.8964E-02 NITH AGE 4 EXITS THE LONER BOUNDARY NITH XN = 0.20136401 VTN = -0.50136-01 VZH = 0.26106-02 0.2304E+01 ™LIN AGE 1 EXITS THE LONER BOUNDARY MITH XN = VYN = -0.7611E-01 VZN = 0.9585E-02 RIGHT YXN = -0.33796+00 YYN = 0.30006+01 YZN = II EXITS THE **B EXITS THE** PARTICLE NO. 210 NITH AGE PARTICLE ND. 499 WITH AGE WITH AGE **NITH AGE** VXN = -0.3448E+00 VON = -0.2406E+01 VXN = -0.6630E+00 VON = -0.4766E+00 VXN = -0.3787E+00 192 221 5 212 215 PARTICLE ND. PARTICLE NO. PARTICLE ND. PARTICLE NO. PARTICLE ND. PARTICLE NO.

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MATICLE M. 576 MITH AGE 11 EXITS THE RIGHT BOUNDARY MITH XM = 0.1031E+02 YM = 0.5098E+02 ZM = 0.1600E+02 VXM = 0.1017E+01 VYM = 0.4196E+01 VZM = 0.1051E-01 0.1356E+02 MATICLE NO. 806 WITH AGE 6 EXITS THE RIGHT BOUNDARY WITH XM = 0.1206E+02 YM = 0.5082E+62 ZM = 0.1214E+02 VXM = 0.8000E-01 VTM = 0.3370E+01 VZM = -0.5235E-02 0.1653E+02 0.1666E+02 0.1340E+02 ZN = 0.1207E+02 0.1191E+02 0.14576+02 IN = 0.40976+01 MATICLE ND. 611 MITH AGE 11 EXITS THE RIGHT BOUNDARY MITH XM = 0.1050E+02 YM = 0.5029E+02 ZM = VXM = -0.2407E+01 VTM = 0.2954E+01 VZM = -0.1226E-01 -73 -MITH AGE 13 EXITS THE RIGHT BOUNDARY MITH XN = 0.1970E+02 YN = 0.5133E+02 VYN = 0.4894E+01 VZN = -0.3297E-02 YN = 0.5061E+02 YN = 0.5202E+02 BOUNDARY WITH XN = 0.1847E+02 YN = 0.5240E+82 TN = 0.5027E+02 TN = 0.5003E+02 PARTICLE ND. 675 NITH AGE 13 EXITS THE RIGHT BOUNDARY NITH XN = 0.1639E+02 VXN = 0.2669E+00 VYN = 0.4530E+01 VZN = -0.2594E-01 WITH AGE 11 EXITS THE RIGHT DOUNDARY WITH XN = 0.1763E40C VYN = 0.4622E401 VZN = -0.3055E-01 PARTICLE ND. 644 NITH AGE 13 EXITS THE RIGHT BOUNDARY NITH XM = 0.1374E462 VXM = -0.1249E401 VYM = 0.3399E401 VZM = -0.3696E-01 MITH AGE 13 EXITS THE RIGHT BOUNDARY MITH XM = 0.7805E+01 VTN = 0.2468E+01 VZN = 0.6095E-02 ARTICLE ND. 443 NUTH AGE 13 EXITS THE RIGHT BOUN Van = 0.52316-01 VTN = 0.43286+01 VZN = 0.65796-01 RIGHT PARTICLE ND. 646 MITH AGE VXN = 0.8462E+00 VXN = -0.2266E+01 VXV = -0.0478E+00 PARTICLE ND. 611 PARTICLE ND. 643 Ż 5 PARTICLE ND. 001 PARTICLE ND. PARTICLE NO.

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MATRIX OF NET CATNODE PARTICLE EMISSION FOR ITERATION NUMBER 75 The cathode is divided into 10 segments in Y AND 10 segments in Z

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R MATRIX OF AVERAGE CATHODE CURRENT DENSITY IN AND'S PER CHAME FOR ITERATION NUMBER The cathode is divided into 10 secrets in Y and 10 secrets in Z

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52.55

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	-	31.26	21.70	19.18	16.48	13.04	16.66	21.13	24.66	23.83	32.25
	•	X. 3	23.33	20.05	19.04	14.67	16.15	20.37	22.39	25.05	29.95
	-	32.30	23.29	19.25	17.02	1.13	16.69	21.96	24.00	24.77	32.74
	•	32.74	22.14	19.25	17.23	14.93	15.32	20.30	23.04	29.62	31.47
	~	32.45	21.05	20.41	17.49	14.42	17.31	20.44	23.11	24.95	31.76
	•	32.56	21.56	19.79	16.40	13.04	15.86	20.76	23.04	24.37	20.02
	•	31.07	20.4	18.87	15.39	11.54	12.15	10.16	20.15	21.78	26.54
	2	37.42	\$1°5	28.91	26.93	27.51	29.24	32.81	30.79	32.66	30.03

## ----EXITING CURRENT FOR ITERATION NUMBER

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1.8169 -1.6478 9.9 9.8 9.8 0.9	BOUNDARY = BOUNDARY =
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NERAGE CURRENT OFF THE CATHODE = 3.9437 INSTANTANEOUS CURRENT OFF THE CATHODE = 3.61	instantaneous cathode current fluctuation square) Vedage cathore current fluctuation saladed =	AVERASE EXITING CURRENT = 3.6639	INSTANTANEOUS EXITING CURRENT - 3.0029	AVERAGE EXITING CURRENT FLUCTUATION SQUARED =	INSTANTANEOUS MIS FLUCTUATION CURRENT RATIOS = AVERAGE PHS FLUCTUATION CURRENT RATIOS = 0.1
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#### NOISE PARAMETERS FOR THE X-NOISE REGIONS

	MUBER OF PARTICLES	TEMPERATURE	TEMPERATURE	CHARGE FLON	CURRENT FLUCTUATIONS	CURRENT RATIOS	CURRENT
8 * 1	0.36335+04	0.6822E+05	0.61872+05	-0.14986404	0.19116+00	0.05405+00	-6. A0146461
ю н н	0.1585E+04	0.6968E+05	0.6455E+05	6.4570E+03	6.3482E-02	0.4925E400	A KKAPADI
•	0.6760E+03	0.6758E+05	0.6753E+05	0.6310E+03	0.4047E-02	0.45616+00	0.3639F401
н) Н	0.6510E+03	0.9946E+05	0.8944E+05	0.6490E+03	0.6306E-02	0.36032+00	6.3590F401
9 n I	0.5740E+03	0.1124E+06	0.1121E+06	0.6450E+03	0.1496E-01	0.26465+00	
1 = 7	0.5640E+03	0.1220E+06	0.1332E+06	0.6560E+03	0.5504E-01	0.3311E+00	0.3484E+D1
• •	0.5760E+03	0.1301E+06	0.1478E+06	E0+30409.0	0.17956-04	0.2578E+00	0.34115+01
6 = H	0.5950E+03	0.1528E+06	0.1557E+06	<b>6.5</b> 820E+03	6.2410E-02	0.25146+00	0.3340F+01
01 = 1	0.5540E+03	0.1670E+06	0.1629E+06	0.5920E+03	0.10436-01	0.3035E+00	0.32456+01
0 = 1	0.5540€+03	0.16705	8	+06 0.1629E+06			

0.31485+01	0.2762E+01 0.2762E+01 0.2612E+01	0.2270E+01 0.2270E+01 0.2074E+01	. 0.1878E+01 0.1668E+01	- 0. 14535+01 0. 12215+01 0. 10025+01	0.7915E+00 0.6040E+00 0.4365F+00	0.29355+00 0.10475+00	0.55336-01 0.22516-01 0.73106-02	0.1996E-02
0.31362+00		0.24395+00 0.23565+00 0.24415+00	0.20002+00	0.19445+00 0.1666E+00 0.1520E+00	0.12596+00 0.10436+00 0.40436+00	0.73525-01 0.57215-01	0.3052E-01 0.2565E-01 0.1390E-01	0.74676-02
0.19516-01	0.1525-05 0.1525-05	0.2007E-01 0.2776E-01 0.8252E-01	0.35576-06	0.142/5-01 0.62705-02 0.73615-02	0.3183E-02 0.2159E-04 8.7799E-09	0.2534E-02 0.1421E-02 0.4414E-03	0.15705-03 0.13042-03 0.27665-05	0.39832-05 0.1106E-06
0.5320E+03 0.4760E+03	0.4960E+03 0.4660E+03	0.4070E+03 0.3720E+03 0.3160E+03	0.3190E+03 0.2940E+03	0.2/002+03 0.2300E+03 0.1620E+03	0.1300E+03 0.1060E+03 0.4200F+03	0.4300E+02 0.2690E+02 0.1500E+02	0.1200E+02 0.6000E+01 0.1000E+01	•••
0.16402+06 0.16802+06	•.15//5400 •.1741E+06 •.1798E+06	0.1096E+06 0.2005E+06 0.2036E+06	0.1909E+06 0.1786E+06	0.19112+00 0.1920E+06 0.2011E+06	0.2128E+06 0.2305E+06 0.2345E+06	0.25136406 0.22596406 0.203776406	0.15156+06 0.4771E+05 0.3503E+05	0.1051E+04
0.1630£+06 0.1700£+06	0.1747E+06 0.1015E+06	0.19315+06 0.21545+06 0.20036+06	0.1947E+06 0.1720E+06	0.17346+06 0.17346+06 0.16246+06	0.2002E+96 0.2188E+96 0.3181E+86	0.3008E+06 0.1593E+06 0.1593E+06	0.1633E+06 0.3967E+01 -0.6825E+01	e. 1 05 9 E + 0 5 e. 0
8.5940E+83 9.6180E+83	0.5440E+03	0.4890E+03 0.4310E+03 0.4200E+03	\$.3420E+03 .3010E+03	e.ce/acres e.1800E+03 e.1440E+03	8.1100E+03 8.6000E+02 8.4900E+32	0.3600E+02 0.9000E+01 0.1400E+02	6.3000E+01 9.1000E+01 9.9	•••
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### NOISE PARAMETERS FOR THE Y-NOTSE REGIONS

a Notse Average Rent Current 1105	••••	.4513E-02 0.2191E-0 .0696E-01 0.2604E+01 .1053E+00 0.4095E+01 .9501E-01 0.5351E+01 0.4512E-01 0.5330E+01		.1003E+00 0.1472E+0 .1255E+00 0.1543E+0 .9530E-01 0.1653E+0 .1776E+00 0.1773E+0 .1376E+00 0.1775E+0 .1330E+00 0.1775E+0 .1330E+00 0.1961E+0 .1330E+00 0.1961E+0 .1151E+00 0.2032E+0 .1151E+00 0.2032E+0 .1151E+00 0.2032E+0
INSTANTANEOUS IN CURRENT CUR FLUCTUATIONS RAY	•••	0.48022-19 0.11295-02 0.19105-05 0.19105-03 0.96315-03 0.96315-03	0.1252E-02 0.3944E-03 0.1439E-03 0.1439E-03 0.7439E-04 0.1946E-02 0.1338E-01 0.1338E-01	6.22395-02 6.37302-02 6.15602-02 6.15602-02 6.35522-03 6.62592-03 6.11572-02 6.11572-02 6.21622-04 6.21622-04 6.21622-04 0.231622-04 0.231622-04 0.231622-04 0.231622-04 0.231622-04 0.231622-04 0.231622-04 0.231622-05 0.231622-04 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.2316222-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.231622-05 0.2316222-05 0.23162200 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222-05 0.2316222000000000000000000000000000000000
INSTANTANEOUS CHARGE FLON	•••	0.0 0.5200E+02 0.7400E+02 0.9200E+02 0.1100E+03	0.1590E+03 0.1750E+03 0.1900E+03 0.2020E+03 0.2260E+03 0.2540E+03 0.2540E+03	0.25206+03 0.26206+03 0.26206+03 0.30106+03 0.30106+03 0.32906+03 0.32906+03 0.35906+03 0.40406+03 0.40406+03 0.40406+03
AVERAGE Temperature	•••	0.10352.06 0.20415.06 0.42665.06 0.51435.06 0.55435.06 0.55435.06	0.6412E+96 0.6721E+96 0.6721E+96 0.6820E+96 0.6806E+96 0.6747E+96	0.6853E+06 0.7130E+06 0.7130E+06 0.71511E+06 0.7992E+06 0.9030E+06 0.9030E+06 0.9030E+06 0.91040E+07 0.11060E+07 0.1106E+07 0.1164E+07
ZNSTANTANEOUS Temperature		0.16005+06 0.20675+06 0.42045+06 0.52745+06 0.52745+06 0.49645+06	0.63065+06 0.66912+06 0.62065+06 0.60695+06 0.73945+06 0.73945+06 0.60132+06	0.7270E+06 0.6776E+06 0.7601E+06 0.7407E+06 0.7407E+06 0.8912E+06 0.8912E+06 0.1043E+07 0.1152E+07 0.1152E+07 0.1179E+07 0.1179E+07
INSTANTANEOUS NUMBER OF PARTICLES	•••	0.0200E+02 0.2050E+03 0.2940E+03 0.2340E+03 0.2540E+03 0.2640E+03	0.2010E05 0.2000E05 0.2960E05 0.2760E05 0.3210E05 0.330E05 0.330E05	6.32400.03 6.3400.03 6.34100.03 6.35600.03 6.35600.03 6.35000.03 6.39300.03 6.39000.03 6.39100.03 6.41900.03 8.41900.03
REGION	8 M 4 1 H H 7 7 7	89769 1111111 777777		2585855558585858 * * * * * * * * * * * * * * * * * * *

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10,72000	0.23495401 0.2565.00			10+31502-0	10+34-62-0	0.3066E+D1	0.31812+01	0.3301E+01	0.3424E+01	0.3558E+01	0.3732E+01	0.3721E+01	0.37146+01	0.17075401		U-3080E+01	0.3663E+01	0.3675E+01	0.2970E+01		AVERAGE CURRENT	0.0	-0.1441E-02	0.18685+00	0.10526+00	0.2827E-01	0.10536-01	0.80%E-02	0.1774E-02	0.2439E-02	-0.22176-03	-9.51005-02	-0.12756-01	-0.37596-01	-0.97796-01	-0.19376+00	8.1774E-02 8.8
	U. 1 3612 + 00		90+20/51.9		0.10406+00	0.1265€+00	0.1462E+00	0.1450E+00	0.1332E+00	0.1386E+00	0.1543€+00	0.1478E+00	0.1344£+00	D. IQAR AD			0.11196+00	0.1610E+00	0.1316E+00		RMS NOISE CURRENT RATIOS	0.0	0.17532-01	0.6260E-01	0.4051E-01	0.3920E-01	0.2434E-01	0.2962E-01	0.2660E-01	0.29195-01	0.25636-01	0.33796-01	0.36156-01	0.33556-01	6.4965E-01	0.5654E-01	0.1361E-01 8.8
	9,5003E-02	V.12105-05 A 45745-05	20-34/61.8	20-36222.0	20-34291.0	0.1702E-04	0.7439E-04	9.2402E-03	0.1792E-02	0.1170E-01	0.1153E-02	0.28865-03	0.1667E-02	6. AA1 9F-05	A KATE-AT		6.7613E-02	0.1496E-02	0.1241E-02		INSTANTANEOUS CURRENT FLUCTUATIONS	••	0.1775E-04	0.7070E-03	0.4150E-02	0.1567E-02	0.2300E-04	0.4072E-03	6.4092E-04	0.3765E-03	0.29535-03	0.2601E-04	0.50356-04	0.4253E-03	0.1701E-02	0.17765-04	0.1712E-03 0.0
	8.41885453 6.44085463	A AREARANT		8.3070C+03	50+30626°0	0.54306+03	0.5610E+03	0.5810E+03	0.59606+03	0.6100E+03	0.6660E+03	.6610E+03	0.6640E+03	ASSAF 401	A LADRANT		0.6670E+03	0.6430E+03	0.5190E+03		INSTANTANEOUS Charge Floh	0.0	-0.1006401	0.3600E+02	0.3000E+02	-0.2000E+01	0.1000E+01	• 5000E+01	0.2000E+01	-0.3000E+01	0.3000E+01		-0.1000E+01	-0.3000E+01	-0.1000E+02	-0.3500E+02	-0.2000E+01 6.6
7012000	101206111 0 10775107		18434751°8			0.15256+07	0.1562E+07	0.1640E+07	B.1700E+07	0.1716E+07	0.1509€+07	0.95496+06	0.0647E+06	AA7 OF LOA	A DIREFAR	90+302/4.0	0.10236407	<ul> <li>1066E+07</li> </ul>	•••		AVERAGE TEMPERATURE		0.883AE+04	0.15356+07	0.1466E+07	0.1451E+07	0.14546+07	0.1446€+07	0.1443E+07	0.1461E+07	0.14496+07	0.14465+07	0.14595+07	0.14485+07	0.1470E+07	0.15385+07	0.0002E+06 0.0
	V.12295479/					0.1503E+07	0.15866+07	0.1630E+07	0.16562+07	0.1701E+07	0.1503E+07	0.93536+06	0.345406	8.8596E+06	A ACCELAN		0.1031E+07	0.1030E+07	••	DISE REGIONS	INSTANT <b>ANEOUS</b> TEMPERATURE		0.0354E+06	0.1561€+07	0.1446E+07	<ul> <li>1435E+07</li> </ul>	0.1444E+07	0.1511E+07	0.1402E+07	0.1472E+07	0.14505+67	0.14505407	0.14816407	0.1404E+07	0.14545+07		0.67542+06 0.0
	6.37165453 6.4176545			CA430074"A		6.46366403	\$.4350E+03	<b>8</b> .4680E+03	0.4600E+03	•.4650E+03	8.4100E+03	e.3350E+03	0.2700E+03	0.27905+03	B. 9590F401	60+36262*A	9.24905+83	0.25886+03	•.•	TERS FOR THE 2-H	INSTANTANEOUS NUMBER OF PARTICLES		0.5410E+03	0.1078E+04	0.1121E+04	0.1097E+04	0.11106+04	0.1151E+04	0.100E+0	0.10715+04		8.1145E404	0.1101E+M	0.11516+04	0.11136404		0.95445493 0.0
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