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# NAVAL POSTGRADUATE SCHOOL Monterey, California



## THESIS

EVALUATION OF  
GEOMAGNETIC ACTIVITY IN THE MAD  
FREQUENCY BAND (.04 to 0.6 Hz)

by

Jeffrey Mark Schweiger

October 1982

Thesis Advisor:

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Evaluation of Geomagnetic Activity in the MAD Frequency Band  
(.04 to 0.6 Hz)

by

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Lieutenant, United States Navy  
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Submitted in partial fulfillment of the  
requirements for the degree of

MASTER OF SCIENCE IN SYSTEMS TECHNOLOGY  
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## ABSTRACT

After defining geomagnetic noise as it applies to MAD, the geomagnetic indices currently used by the fleet to predict MAD geomagnetic noise are reviewed to determine their actual applicability. The current indices are determined to be insufficient, methods are proposed for establishing a new MAD index, and a developmental MAD index system was tested. Geomagnetic fluctuations in the .04 to 2.0 Hz frequency band were recorded at Monterey, California, and used for a preliminary test of the proposed MAD index.

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## I. GEOMAGNETICS REVIEW

### A. HISTORY OF GEOMAGNETICS

The beginnings of the study of geomagnetism lie back in prehistory when magnetic attraction between iron and certain minerals was first observed. Exactly when this phenomenon was first noticed is not known, but the properties of magnetite, then called lodestone, appeared in Greek literature around 600 B. C. (Brennan and Davis). [Ref. 1]

Chapman [Ref. 2] indicates that the directional property of magnets was known and used in Europe prior to 1200 A. D. and possibly in China before then. E. N. Parker [Ref. 3] noted "It is an interesting fact that the ancient walls of Peking were lined up with magnetic north rather than geographic north, a difference at that time of about 10. We may presume that the surveyor found it easier to work with his compass needle by day than to sight on the pole star by night." This property allows the use of the Earth's geomagnetic field for navigational purposes.

By the mid-fifteenth century it was determined in Europe that the magnetic compass does not point to true north. The angle between true north and the direction indicated by the compass is now known as magnetic declination by the geophysicist and as variation by the navigator.

The magnetic field dip, or magnetic inclination is the angle, in a vertical plane, between the horizontal and the direction of the earth's magnetic field vector. It was observed in 1544 by an instrument maker in Nuremberg named Hartmann, and again by Robert Norman in London in 1581.

These discoveries or observations gave rise to the study of geomagnetics as a specialty which gained its cornerstone

with William Gilbert in 1600. After comparing his experimental results with the previous work of others such as Norman, Gilbert, in his book, "De Magnete," concluded that "Magnus magnes ipse est globus terrestris (the earth globe itself is a great magnet)" [Ref. 2]. It is this concept, that the earth is itself a magnet, that is the basis of the science of geomagnetics.

Gilbert felt that the Earth's magnetism must remain constant except for geological changes, but it was soon determined that this was not the case. A 'secular variation' of the Earth's was found to exist.

Shorter term changes in the geomagnetic field were observed and it was eventually realized that geomagnetism is dynamic. In 1722 George Graham discovered that daily, or diurnal, variations exist [Ref. 2]. During the early Nineteenth century magnetic observatories began to be established to record the changes in the geomagnetic field in a systematic fashion (Knecht) [Ref. 4].

## B. EARTH'S MAGNETIC FIELD

### 1. Constituents of the Geomagnetic Field

There are various ways of breaking down the constituting parts of the geomagnetic field. One way is to divide the field in terms of distance from the center of the earth. Doing this yields these three parts: internal, crustal, and external (AFGL) [Ref. 5]. The internal field originates in the core region and is the more stable field, containing only extremely low frequency temporal variations. The crustal (or anomalous) field arises from modifications made on the internal field by materials and structures in the Earth's crust. These variations are not spatially constant and give rise to some of what is known as geological

variations. The external field is the most dynamic and arises from many sources including the interaction between the solar wind and the terrestrial magnetic field.

Another way of describing the components of the geomagnetic field is by time variation. This division is accomplished by considering that part of the field which varies with periodicities greater than about one year as the steady field and what is left as the variation field (Knecht). [Ref. 4]

The steady field consists of the above named internal field, also referred to as the main field. Slow variations in the main field with periods of years or longer are referred to as secular variations.

## 2. Models of the Main Field

Various models of the main geomagnetic field make use of a geocentric dipole. Gauss, in 1839, demonstrated that, as a fairly good first approximation to the geomagnetic field, the field of a uniformly magnetized sphere (for points outside the sphere) is equivalent to the field of a magnetic dipole located at the center of the sphere (Jacobs). [Ref. 6]

The simplest of the present approximations of the geomagnetic field is that of a short bar magnet or dipole located at the center of the earth with an axis inclined approximately  $11.5^\circ$  from the Earth's axis of rotation. The sense of the field lines is from south to north (Figure 1.1).

The axis of this dipole intersects the earth at the geomagnetic north pole,  $78.5^\circ\text{N}$ ,  $291.0^\circ\text{E}$  in geographic coordinates, and at the geomagnetic south pole,  $78.5^\circ\text{S}$ ,  $110.0^\circ\text{E}$ . The moment of the geomagnetic dipole is  $8.1 \times 10^{22}$  amp-m<sup>2</sup>. It is these poles that are used to define the geomagnetic

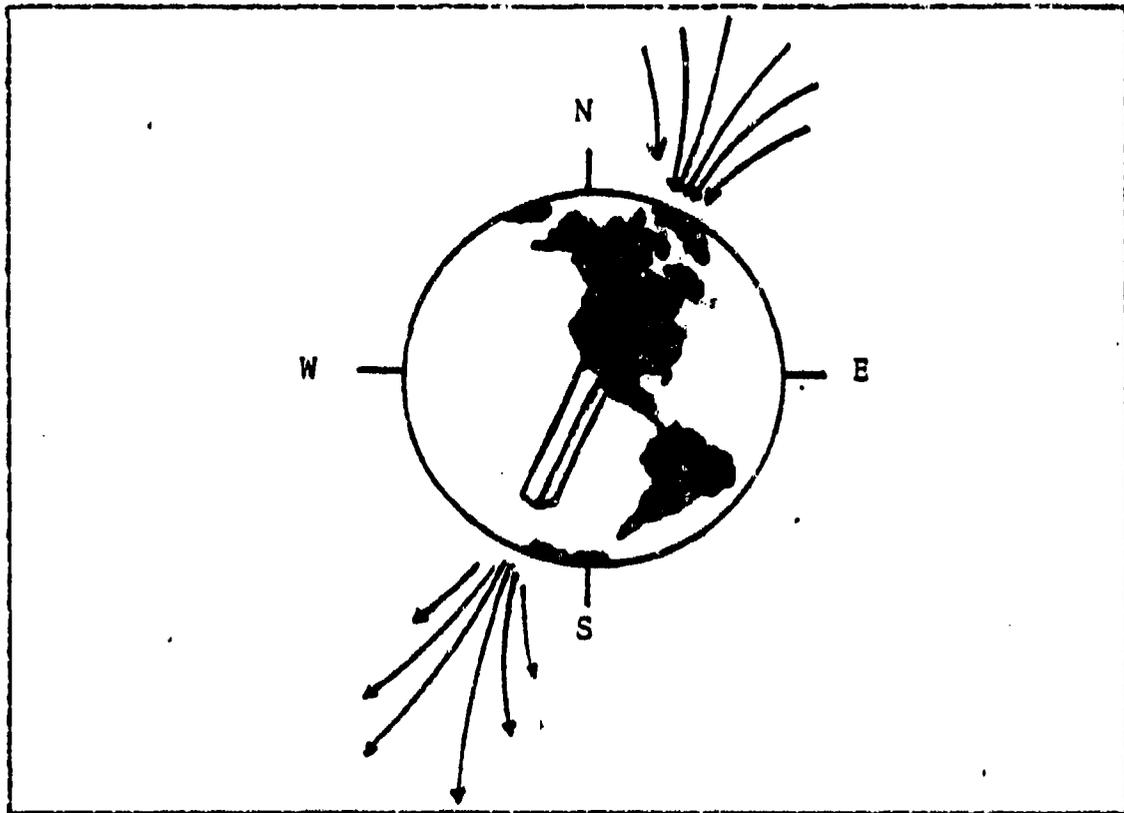


Figure 1.1 Dipole Appearance of Geomagnetic Field

coordinate system (Knecht) [Ref. 4]. The geomagnetic coordinate system is a spherical polar system similar to the geographic coordinate system with a geomagnetic equator defined 90 degrees away from either geomagnetic pole in latitude. This tilted geocentric dipole model describes the main geomagnetic field to an accuracy of about 10%.

In 1940 Chapman and Bartels defined an off-center dipole in the earth's interior, called the eccentric dipole. This dipole is displaced 0.0685 earth radii (436 km) in magnitude from the center and in the direction of the point 15.6°N, 150.9°E (geographic coordinates) (Vestine) [Ref. 7].

The intersections of the eccentric dipole axis at the earth's surface are  $81.0^{\circ}\text{N}$ ,  $84.7^{\circ}\text{W}$  and  $75.0^{\circ}\text{S}$ ,  $120.4^{\circ}\text{E}$  (Figure 1.2) (Haynes) [Ref. 8]. This approximation is accurate to within a few percent.

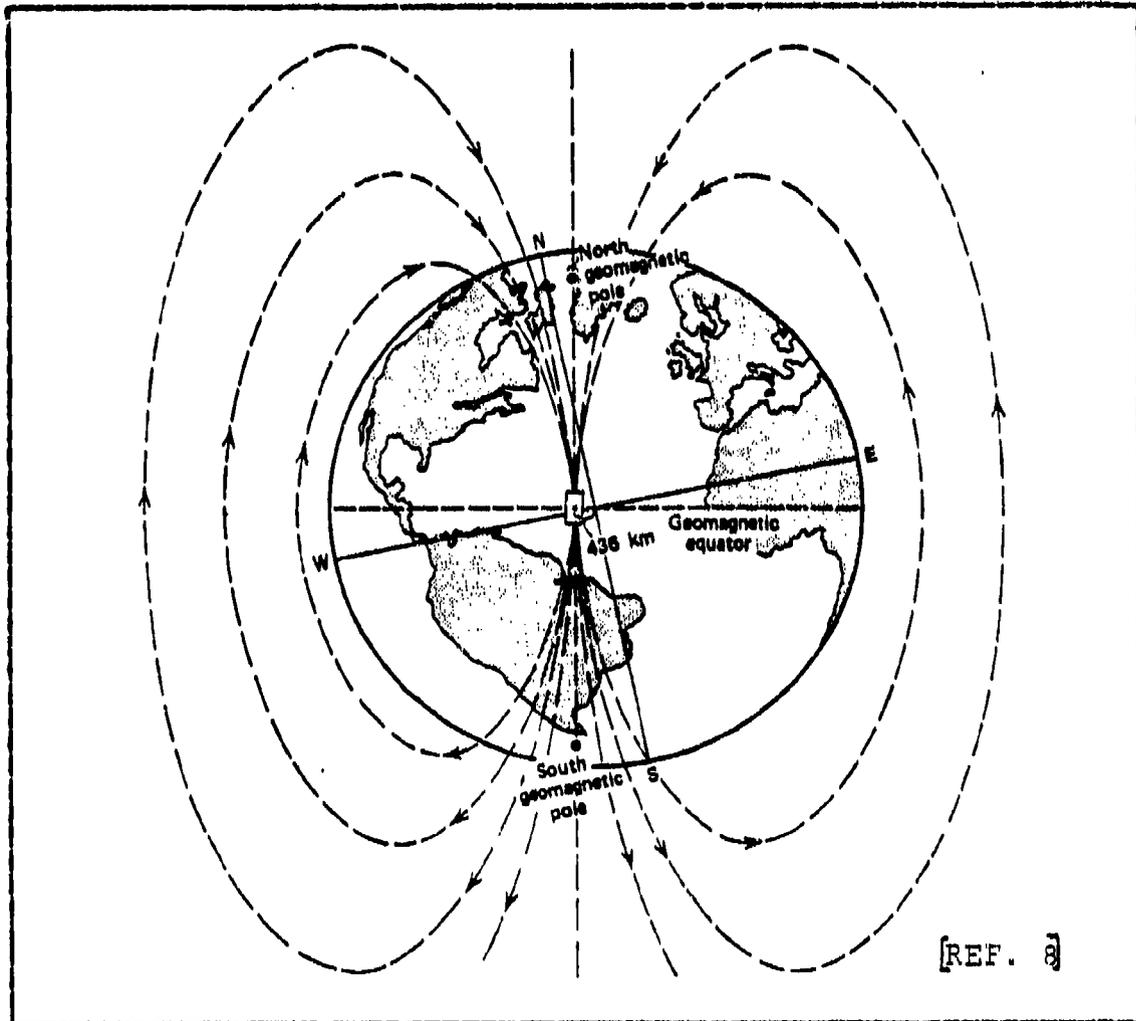


Figure 1.2 Eccentric Dipole Model of Geomagnetic Field

The field has additionally been modeled to an accuracy of about 1% by determining Gaussian coefficients by a least-squares fit of experimental measurements of the

geomagnetic field. These coefficients are used in a spherical harmonic series representing the scalar potential of the field. This accuracy implies that the internal contribution to the total main field is at least on the order of 99%.

The International Geomagnetic Reference Field (IGRF) yields values which differ by only parts per thousand from measured values.

### 3. Sources of the Geomagnetic Field

There are various elements that contribute to the geomagnetic field, some external to the earth's surface and some internal. As previously mentioned, the external contributions make up only a small fraction of the steady field, playing a more important role in the variation field.

These external sources include current systems in the earth's upper atmosphere affected by solar electromagnetic radiation and gravitation, solar corpuscular radiation or the interaction of solar plasma with the main field, and the effect of the solar interplanetary field. [Ref. 4]

Various magnetic surveys of the world, including those conducted at ground level, by airborne instruments, and by satellite, have pointed to the fact that the largest source of the earth's magnetic field is internal to it. While there exists residual permanent magnetism in the earth's crust, this cannot be the principal internal source of the geomagnetic field due to temperature and material properties known to exist in the earth's interior (Nagata and Ozima). [Ref. 9]

Permanent magnetism is generated by microscopic electric currents, since a changing electric field will generate a magnetic field. Another way to generate a magnetic field is by the motion of electric charges in a

macroscopic current. Convective motion of the electrically conducting fluid core of the earth, resulting in a macroscopic current system, is considered to be the principal source of the main field.

The most promising present theory of the generation of the geomagnetic field is that of some sort of a self-exciting dynamo system. This means that the motion of a conductor, such as the molten iron in the earth's core, in a magnetic field produces a current which in turn induces a magnetic field in support of the original magnetic field [Ref. 4, 7]. A very simple model of such a dynamo is shown in Figure 1.3.

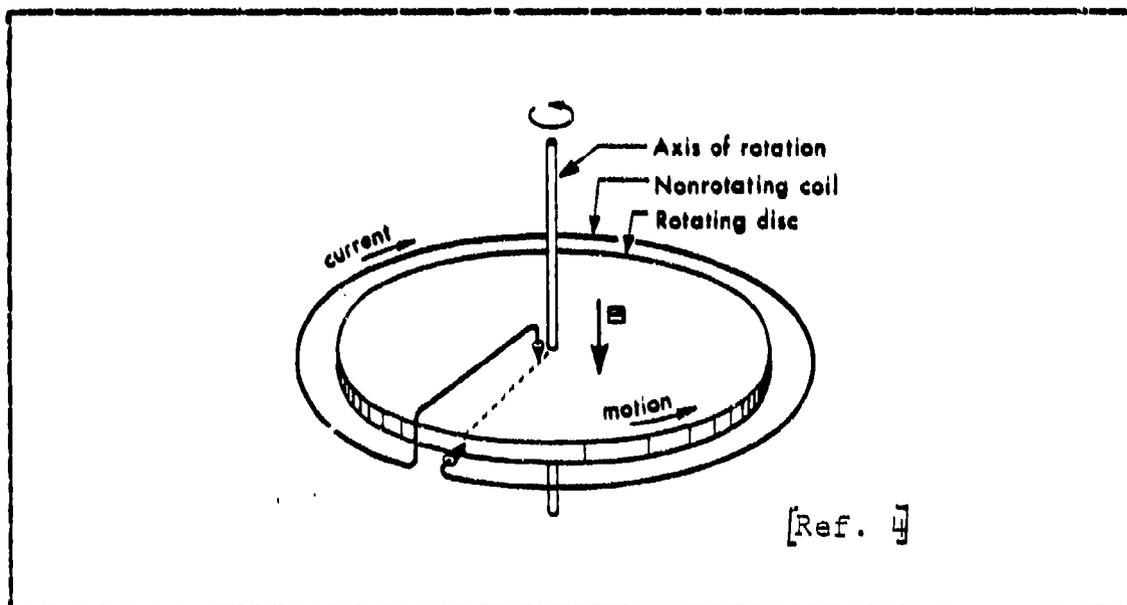


Figure 1.3 Simple Disk Dynamo

The original excitation or seed field may be due to an external field line, perhaps from the solar interplanetary magnetic field. This original, poloidal, field line is wound up due to the differential rotation (rotation not

constant with latitude) in the molten core. The wound up line becomes an intense azimuthal field which is carried outward by the upwelling associated with convection and twisted by the action of the Coriolis force. The twisting generates a helical toroidal field which, by outward diffusion, generates the externally observed quasidipole geomagnetic field.

The combination and interaction of two (see Figure 1.4) or more disk dynamos can also explain the reversal of the geomagnetic field [Ref. 7, 9].

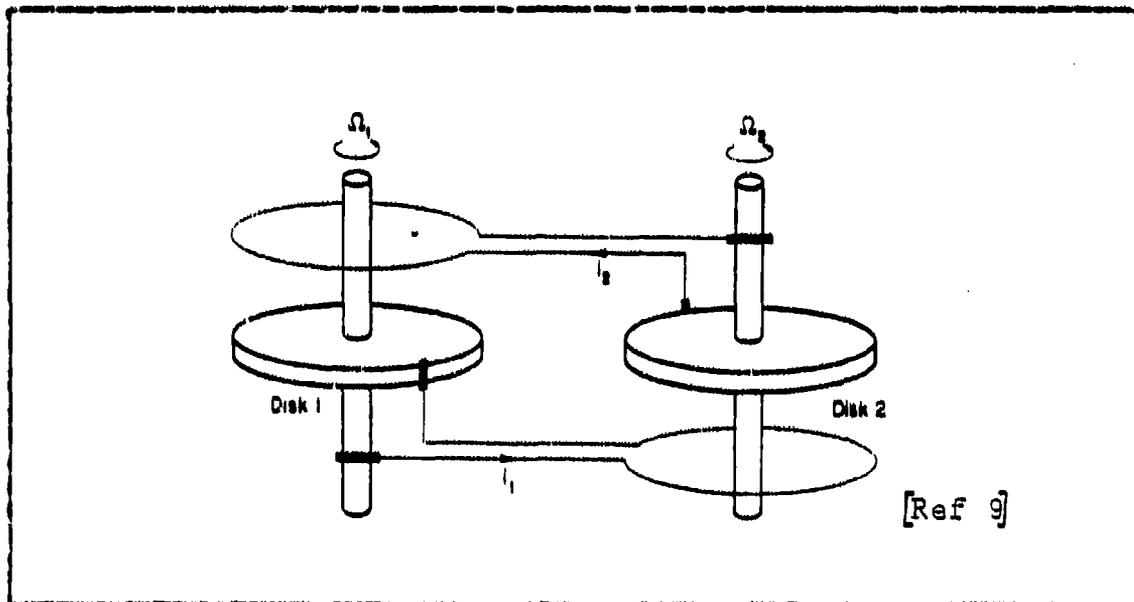


Figure 1.4 Twin Disk Dynamo

Regional anomalies that are nondipolar (do not conform to the dipole field) possibly arise from eddy circulations in the outer core [Ref. 4].

#### 4. Magnetosphere

The magnetosphere can be defined to be that region (see Figure 1.5) occupied by the geomagnetic field above the ionosphere, a region where the field strongly influences the dynamics of ionized gas and charged particles (Kern) [Ref. 10].

If the space surrounding the earth were a perfect vacuum, the earth's magnetosphere and magnetic field might be more or less symmetric and extend outward until it merged with, and its strength became insignificant compared to the solar and other planetary magnetic fields. This turns out not to be the case.

Instead of being an island in a perfect vacuum, the earth encounters a continuous flow of hot, highly conductive ionized gas, or plasma, streaming outward from the sun. This continuous stream of charged particles is called the solar wind. The density of this 'wind' near earth is on the order of 10 particles per  $\text{cm}^3$ , and has a velocity averaging 300-500 km per second (Jacobs). [Ref. 6]

Both the solar wind and the geomagnetic field exert pressure. The hot plasma of the solar wind pushes against the geomagnetic field deforming the field. At distances greater than about 13 or 14 earth radii, the pressure of the solar wind greatly exceeds that of the geomagnetic field and the geomagnetic field will be swept along with the wind. From 8 to 10 earth radii inward, the geomagnetic field pressure will predominate, excluding the solar wind, this being the region of the magnetosphere.

In the intermediate region, the magnitude of the solar wind and geomagnetic field pressures are comparable and the solar wind is compressed and flows around the geomagnetic field. This occurs when the magnetic energy density ahead of the plasma equals the kinetic energy

density of the streaming plasma. The solar wind is stopped at this point and forced to flow around the magnetosphere. This region where the magnetosphere starts is called the magnetopause. [Ref. 8]

The velocity of the undisturbed solar wind is analogous to a 'supersonic' velocity. Thus a shock front is formed between the magnetopause and the solar wind. The magnetosheath is the region of severe turbulence that exists between the shock front and the magnetopause.

Since the solar wind always travels outward from the sun, the effect of the wind on the earth's main field is not completely symmetric, although it is almost symmetric about an axis through the earth and sun. A geomagnetic tail is formed where the wind sweeps the geomagnetic field along with it on the nightside of the earth [Ref. 4]. Figure 1.5 pictorially represents the effects of the solar wind on the geomagnetic field.

## 5. Time Variations of the Geomagnetic Field

The geomagnetic field changes with time. As previously mentioned, very slow variations in the main field with periods on the order of years to thousands of years are referred to as secular variations. Secular variations are geologic or 'paleomagnetic' in origin. Secular variations are not caused by a strength or orientation change of the center dipole. Paleomagnetic studies are used to determine the secular variation. Geologic structure, especially conductivity structure, may partially mask the secular variation at one point on the earth as compared to that at another point.

Other time variations of the field can be categorized into quiet variation fields and disturbed variation fields. Disturbed variation fields include geomagnetic

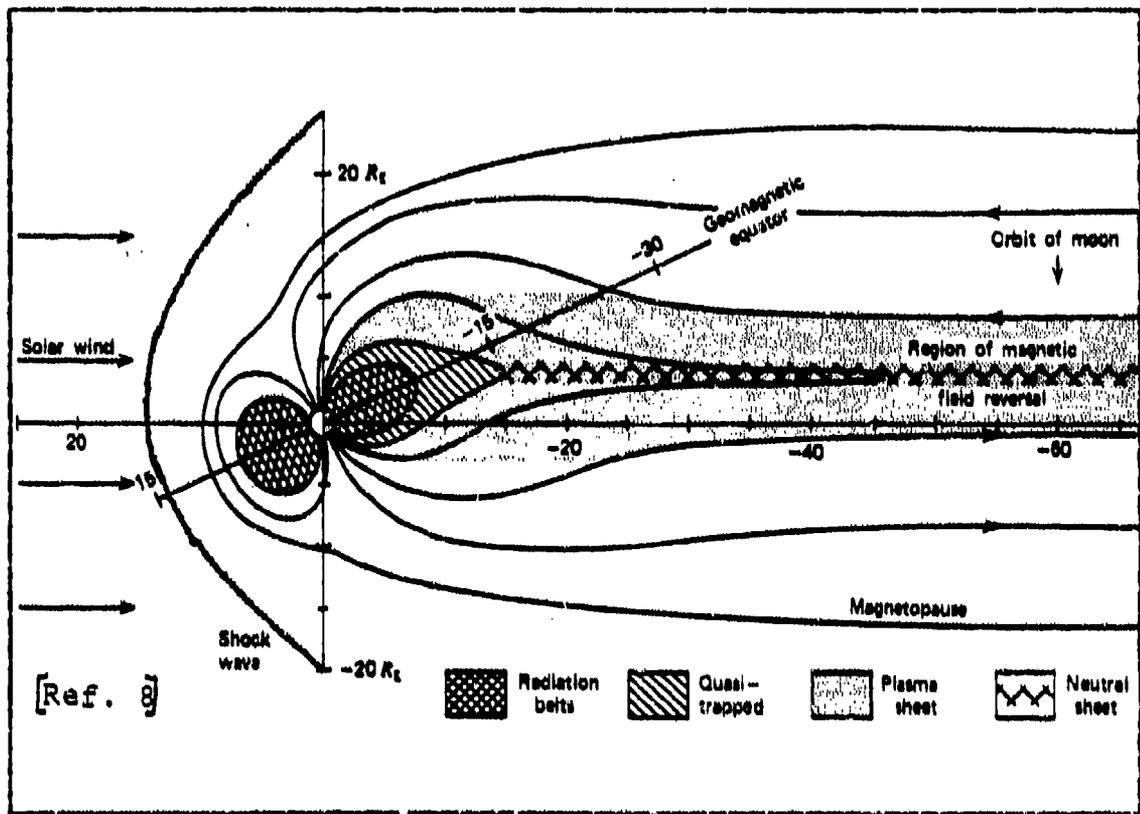


Figure 1.5 Configuration of The Magnetosphere

micropulsations which will be discussed separately, since they are of particular interest as a noise source for MAD sensors.

a. Quiet Variation Fields

Quiet variation fields are those which are not due to disturbances in the interplanetary environment and which vary slowly and regularly [Ref. 4].

There are several contributing fields to the quiet variation. These include the Solar Quiet Daily Variation (Sq), the Lunar Daily Variation (L), and the daily variation due to magnetospheric effects.

The Solar Quiet (Sq) variation is the name given to the pattern of diurnal field variation with respect to solar local time which is caused by currents flowing in the ionosphere (Matsushita) [Ref. 11]. The major portion (about two-thirds) of the Sq field is due to what is referred as an atmospheric dynamo. High speed tidal winds are generated by solar heating causing convection of the upper atmosphere [Ref. 4]. These winds produce a stationary current system by moving the conducting particles of the upper atmosphere across the geomagnetic field lines. The daily variation is caused by the earth rotating under the current system. The remaining third of the Sq variation is caused by currents in the earth induced by the primary currents in the ionosphere.

The Sq field can be shown to be latitude dependent reaching a maximum at the magnetic equator where a concentration of current, the equatorial electrojet, exists [Ref. 4]. The maximum horizontal component intensity is about 100 nT at the equator with 25 to 50 nT more likely at higher latitudes.

Longitudinal, seasonal, and solar cycle dependencies also occur for the Sq field.

An example of the quiet-day variation at Monterey, California is shown in Figure 1.6 and is summarized in Table I. This data was taken using a Cesium Vapor total field magnetometer in February, 1979.

The Lunar Daily Variation, L, is approximately one-tenth the magnitude of the Sq field and exhibits a semi-diurnal behavior in lunar time [Ref. 4]. The major difference is that the winds are caused by lunar-solar gravitational tides. The L field is dependent on seasonal influences, lunar phase, the solar cycle, and latitude.

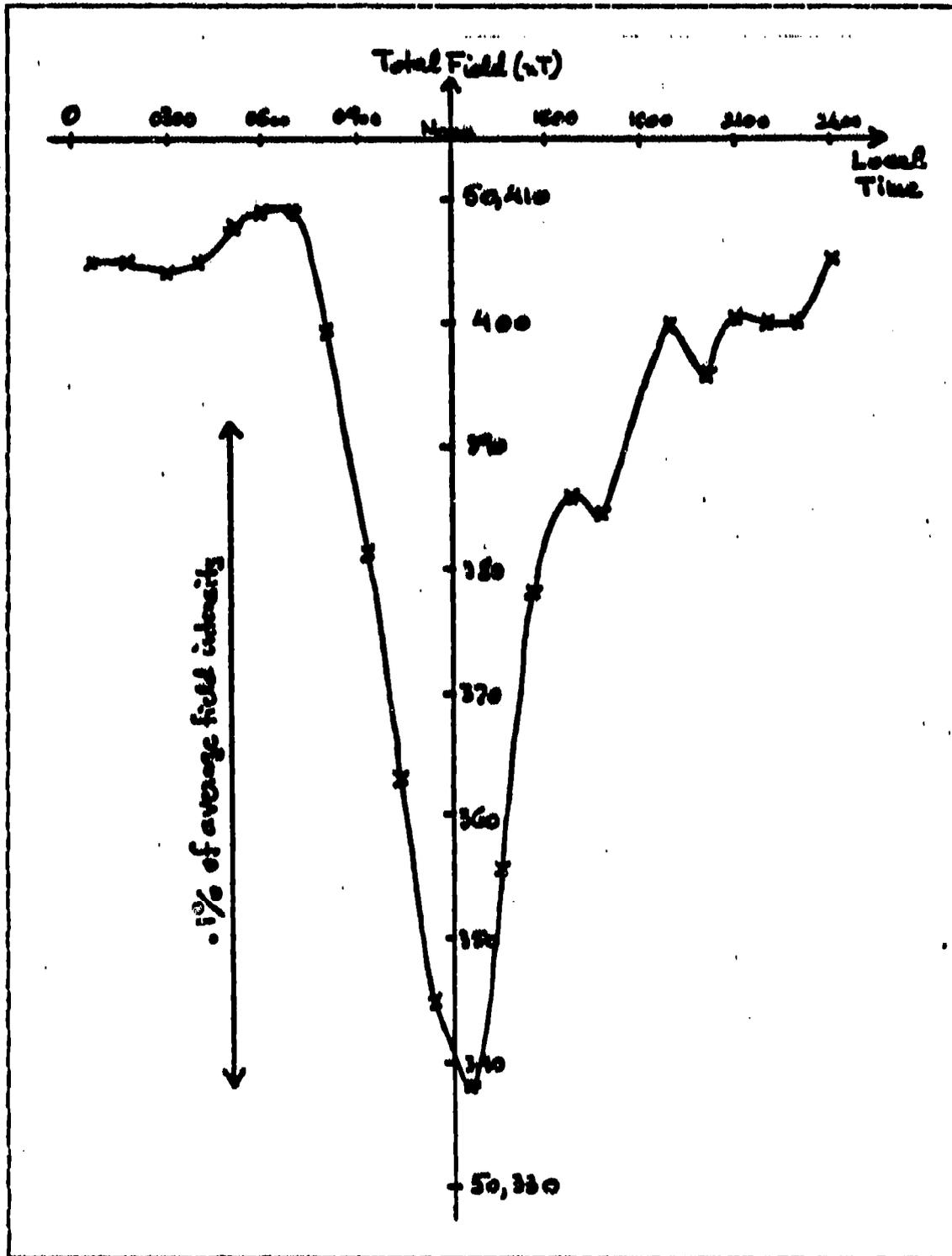


Figure 1.6 Variation of Total Geomagnetic Field Intensity, February 26, 1979

TABLE I

Quiet Day Variation in La Mesa Village, February, 26, 1979

POSITION 36°36'N, 121°51'W (LA MESA VILLAGE)

CALCULATED VALUES USING THE 1975 U.S. CHART MODEL (WORLD DATA CENTER A, BOULDER, COLORADO) FOR FEBRUARY 1979:

	D	I	H(NT)	Z(NT)	F(NT)
VALUE (FEB. 1979):	15.96°	60.74°	24,650	44,007	50,441
YEARLY CHANGE :	-2.3 '	-.8 '	-4.3	-32.1	-30.1

MEASURED VALUES OF TOTAL FIELD INTENSITY FOR FEB. 25, 1979

DATE	TIME	F(NT)	
2/25/79	00:44	50,405.24	
	01:48	50,405.79	
	02:52	50,404.80	
	03:56	50,405.48	
	05:00	50,408.19	
	06:04	50,409.15	
	07:08	50,409.66	
	08:12	50,399.69	
	09:16	50,381.59	
	10:20	50,363.68	
	11:24	50,345.08	
	12:28	50,338.51	MID-DAY LOW
	13:32	50,356.48	
	14:36	50,378.86	
	15:40	50,386.98	
	16:44	50,384.30	
	18:51	50,400.81	
19:55	50,396.91		
20:59	50,401.85		
22:03	50,400.80		
23:07	50,400.54		
2/26/79	00:11	50,406.51	

AVERAGE 50,390.50 ± 21.43 (NT)

$$F_{\text{CALC}} - F_{\text{MEAS}} = 50.5 \text{ NT OR } .1\%$$

A diurnal effect due to the dayside-nightside difference in compression by the solar wind of the geomagnetic field causes a small variation of the order of 3 nT [Ref. 4].

#### b. Disturbed Variation Fields

Disturbed variation fields are geomagnetic field variations that appear to be the result of interplanetary environmental changes and do not possess a simple periodicity. These variations include ionospheric disturbances, the aurora, geomagnetic storms, and geomagnetic micropulsations.

An ionospheric disturbance is a departure from the normal behavior of the ionosphere.

Solar flare effects (SFE) are magnetic disturbances produced by X-rays emitted from the solar flare. SFE's usually have a rapid onset, typically a few minutes in duration, followed by a slower return to normal. The entire event lasts on the order of an hour (Reid). [Ref. 12]

Auroras are caused by the precipitation of charged particles down magnetic field lines into the atmosphere and can be one of the brightest visual phenomena in the sky. The more intense and active auroras occur with geomagnetic disturbances and greatly increase ionization as well as creating the spectacular visual displays. [Ref. 4]

Geomagnetic storms are due to a change in the dynamic pressure of the solar wind. A typical storm begins with a compression of the magnetic field by an increase in solar wind dynamic pressure called a sudden commencement (SC), which increases the magnetic field (The so-called "gradual storm" begins with a gradual increase in field strength). The increase in field strength is on the order of several tens of nanoTeslas (nT) and takes about one to

six minutes to rise. If a disturbance starts with an SC but lacks the succeeding stages of a storm it is referred to as a sudden impulse (SI). Following the SC, the field remains compressed for two to eight hours in the initial phase of the storm. The main phase follows the initial phase. Over a period of hours to a day a westward ring current is set up at a distance of several earth radii whose magnetic field leads to a decrease in field strength on the order of 100 nT. This decrease overshoots the equilibrium field strength and leads into the recovery phase of a day or longer where the field returns towards its prestorm strength as the ring currents gradually dissipate [Ref. 4], (Matsushita) [Ref. 13]. A magnitude-time graph of a typical geomagnetic storm is shown in Figure 1.7.

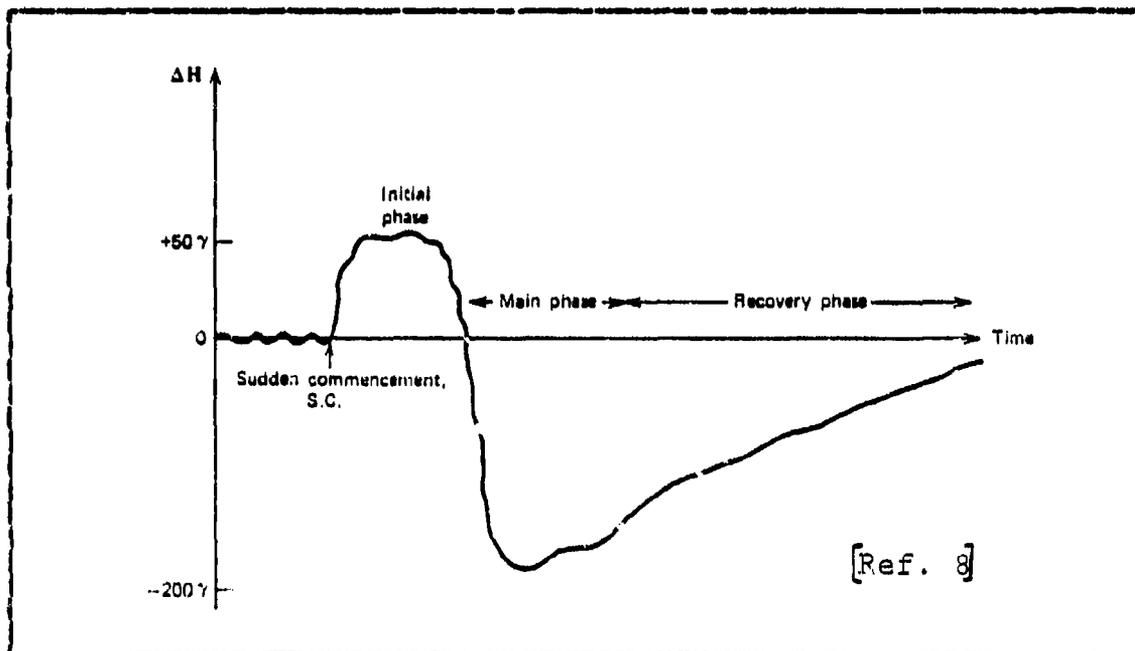


Figure 1.7 Typical Midlatitude Geomagnetic Storm

**Geomagnetic micropulsations** are rapid fluctuations in the surface magnetic field with periods of about 0.2 seconds to 10 minutes (frequencies about 0.0016 to 5.0 Hz). These are observed as a type of geomagnetic disturbance by ground based magnetometers [Ref. 4]. Micropulsations will be discussed in depth later.

## 6. Elements of the Magnetic Field Vector

The geomagnetic field vector is measured or characterized at any point by its direction and magnitude. This can be done in terms of some set of three independent parameters such as two direction angles and the magnitude, or three perpendicular components. [Ref. 4]

The system of coordinates commonly employed for describing the geomagnetic field on the surface of the earth is shown in Figure 1.8. The field is measured in terms of local (geodetic) coordinates with respect to True North.

The various coordinates are referred to as magnetic elements and are defined as follows:

**B:** Total Field Intensity (the symbol  $F$  is also used)

**H:** Horizontal Component

**X:** Northward, or North-South Component

**Y:** Eastward, or East-West Component

**Z:** Downward, or Vertical Component

**D:** Declination or magnetic variation. This is the angle between **X** and **H** and is measured positive eastward.

**I:** Inclination or dip angle. This is the angle between **H** and **B** and is measured positive downward.

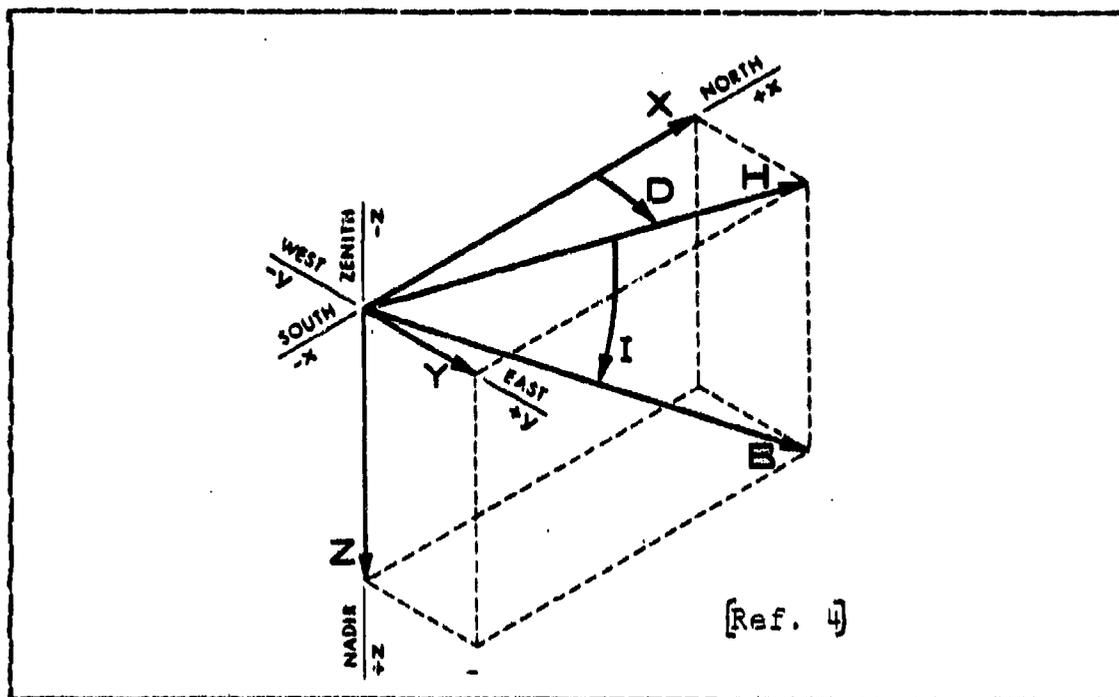


Figure 1.8 Magnetic Field Elements

## II. INTRODUCTION TO MAGNETIC ANOMALY DETECTION (MAD)

### A. DEFINITION OF A MAGNETIC ANOMALY

A magnetic anomaly is defined as any spatial variation or disturbance in the geomagnetic field which is due to local causes. Anomalies can be caused by waves, ore deposits, sea mounts, and magnetized objects such as surface ships and submarines (Anderson) [Ref. 14]. For the purposes of this research, magnetic anomalies due to a simple dipole field, such as those generated by submarines, will be regarded as signals, while other anomalies will be regarded as noise and will be discussed later.

### B. HISTORICAL DEVELOPMENT OF THE MAGNETIC DETECTION OF SUBMARINES

#### 1. Early Detection Systems

Attempts at finding submerged submarines by sensing disturbances in the geomagnetic field date back as least as far as World War I. In 1918, Earnest Merritt, at the Naval Experimental Station, New London studied the use of a fixed coil type of detector for use in moving boats and airplanes (Slichter). [Ref. 15]

MAD, originally known as Magnetic Airborne Detector, and now called Magnetic Anomaly Detection, had its beginnings as an airborne ASW sensor in late 1940 and early 1941.

Anomalies in the geomagnetic field caused by the presence of submarines, are on the order of one to a few nT (gammas) in magnitude. This is quite small compared to the magnitude of the field itself (30000 - 60000 nT). Two methods are generally employed to measure this small

disturbance. One is to use a gradiometer which measures the spatial rate of change of the magnetic field or its gradient. The second method, and the one presently used by U.S. Navy aircraft is to use a magnetometer to directly measure changes in the magnitude of the magnetic field.

The British investigated the use of a gradiometer for submarine detection, and by early 1941, had developed a two-coil gradiometer system which could detect a submarine at a range of 200 feet under favorable conditions. This range was considered to be too small to be of operational value and work on such a system was terminated when a magnetometer system showed promise (Coleman). [Ref. 16]

By late 1940, Victor V. Vacquier of the Gulf Research and Development Company had developed a sensitive saturable core magnetometer intended for geophysical (mineral) prospecting. The Vacquier magnetometer became the basis for further MAD development. The Airborne Instrument Laboratory of Columbia University continued the investigation of means of localizing submerged submarines by MAD.

The simplest saturable core or fluxgate magnetometer consists of a saturable or ferromagnetic core around which a coil of wire is wrapped (see Figure 2.1). This coil carries a sinusoidal current,  $I(t)$ , which is large enough to saturate the core during part of each cycle. The inductance of the coil will change as a function of the magnetization of the core. The core magnetization, in turn, depends on the instantaneous current in the coil, and, if present, the external magnetic field.

In the absence of an externally applied magnetic field, magnetization as a function of exciting current is symmetric around  $I=0$  (Figure 2.2). An external magnetic field parallel to the core's axis will change the magnetization of the core and shift the magnetization curve. This

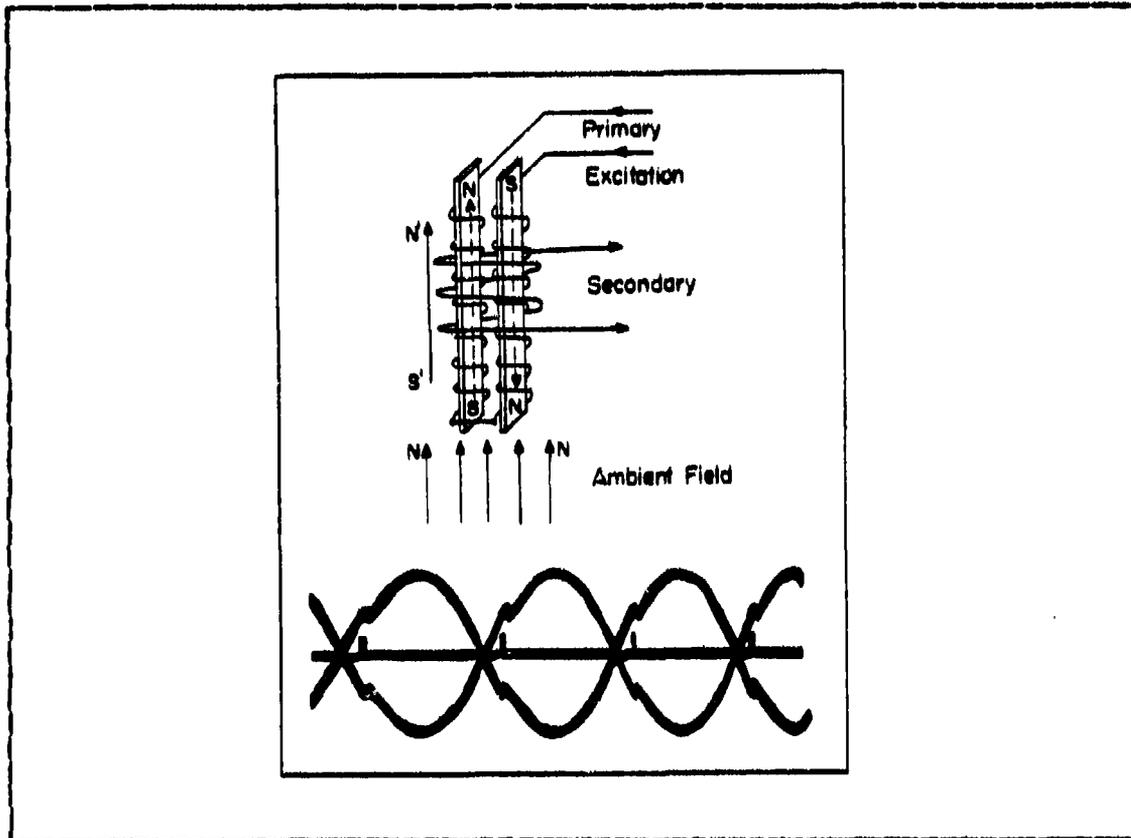


Figure 2.1 Element of Fluxgate Magnetometer

change will cause an asymmetry which can be sensed by analyzing the harmonic content of the signal. The coil can be combined with a stabilizer system, which keeps the detector element (coil) aligned with the geomagnetic field.

This was the basis of the MAD magnetometers used in World War II such as the AN/ASQ-1, ASQ-1A, and ASQ-2 [Ref. 16].

The fluxgate magnetometer measures only the component of the external field parallel to the axis of the ferromagnetic core. In order to measure the total field in this fashion it is necessary to align the ferromagnetic core along the earth's magnetic field or by using mutually

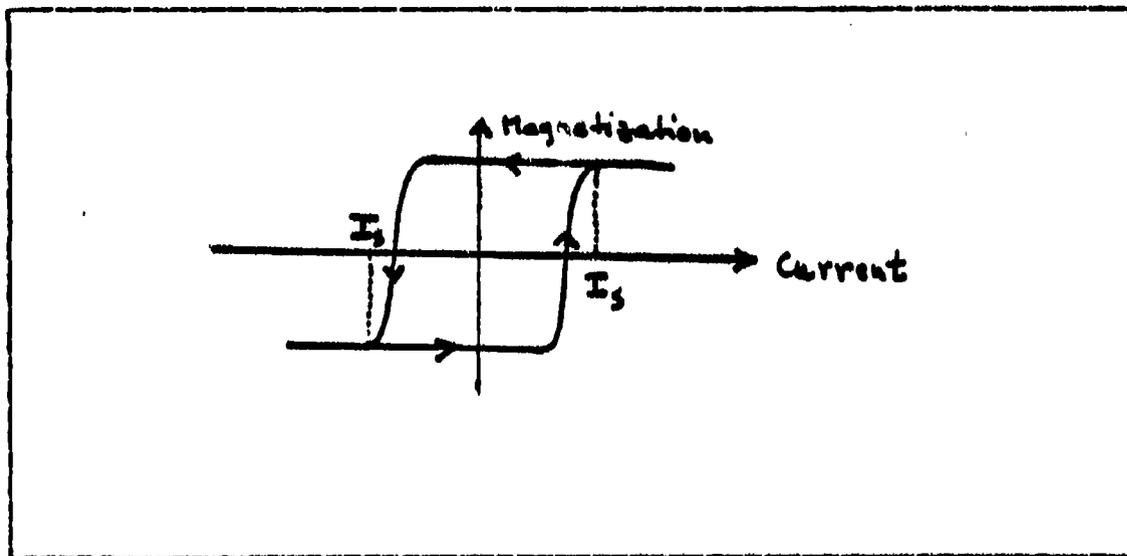


Figure 2.2 Magnetization Versus Exciting Current

perpendicular (orthogonal) cores. Precision requirements made this type of system difficult to realize during World War II but it has since found many years of operational usage and has found widespread use in geophysical exploration work and satellite mapping of the geomagnetic field.

## 2. Early Operational Usage

During World War II MAD provided a passive method of detection and tracking of submerged submarines. Within its range (then on the order of 500 feet) MAD gave a measure of surprise to the attacking aircrew. Until the first attack was delivered the crew of the submerged submarine might not even be aware of the aircraft hunting it.

Operational deployment of MAD began on a limited basis in December, 1941 with the installation of the early Mark I MAD in a blimp at Naval Air Station Lakehurst, New Jersey [Ref. 16].

By the end of 1942, MAD was operational on board PBY Catalina aircraft, nicknamed "Madcats." Even though MAD was operational at this time, it was not until February, 1944 that an initial contact by MAD led to the sinking of a submarine.

During February, 1944, the Madcats of Patrol Squadron 63 were assigned to fly a MAD barrier patrol of the Straits of Gibraltar. On 24 February 1944, a Madcat of VP-63 detected U-761 by use of MAD and commenced tracking to confirm the contact as a moving target. An attack was conducted in conjunction with another Catalina, two destroyers, and eventually two other aircraft, and U-761 was sunk. U-392 was sunk after a similar MAD contact on 16 March, and on 5 May, the third successful attack on a U-boat resulting from an initial MAD contact took place when U-731 was sunk (OEG No. 51, Price). [Ref. 17, 18]

The attacks at the Straits of Gibraltar demonstrated that though MAD has a limited search rate, there are scenarios where it can be employed effectively as a search sensor, such as providing a blockade across a restricted area without the presence of surface craft (OEG No. 54) [Ref. 19].

### 3. Current Systems

The magnetometer system in current operational use is the optically pumped magnetometer. The optically pumped magnetometer measures the external magnetic field by making use of the fact that when an atom is immersed in a magnetic field, its energy levels are split. This is known as the Zeeman effect. For the fields of interest, the amount of splitting of the levels is proportional to the intensity of the magnetic field. By measuring the separation between the levels the magnitude of the magnetic field can be determined.

This type of magnetometer usually makes use of Cesium or Rubidium vapors, or Helium gas. The current operational MAD system, the AN/ASQ-81, is a Helium gas optically pumped magnetometer.

In an optically pumped magnetometer (see Figure 2.3), the sample vapor or gas (such as Helium) is collected into an absorption cell. Circularly polarized light is passed through the cell giving up some of its energy in exciting or pumping the electrons of the sample gas to higher energy levels. These electrons then fluoresce to lower, metastable energy states. This is 'optical pumping.' A detector monitors the degree of optical pumping by measuring the transparency of the gas cell.

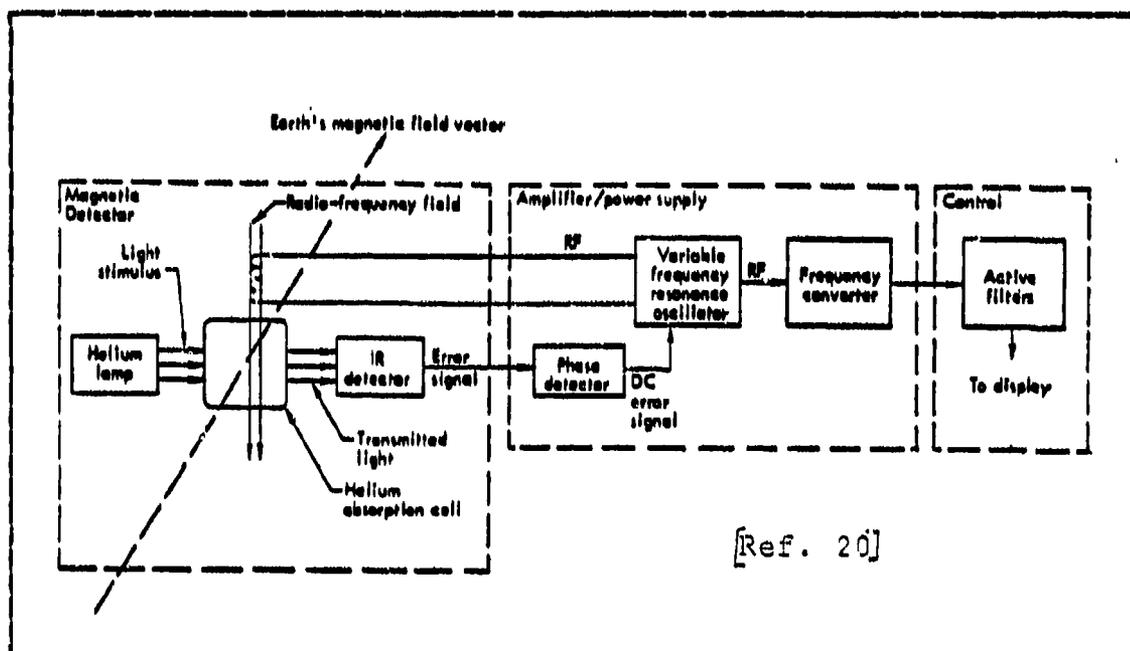


Figure 2.3 Metastable Helium Magnetometer

The actual separation of the energy levels is measured by applying a weak R. F. field which redistributes the

electrons among the ground state sublevels. These transitions will only occur when the R. F. field has a particular frequency (called the Larmor frequency) which is proportional to the separation between the levels and therefore to the external magnetic field [Ref. 20]. Thus measurement of this frequency yields a direct value of the magnetic field.

#### 4. Future Developments

Optically pumped magnetometers are the most sensitive sensor for fleet usage today. Increasing capability to make use of the phenomenon of superconductivity will yield more sensitive MAD sensors. Superconducting Quantum Interference Devices (SQUIDS) which make use of the Josephson effect have a theoretical sensitivity of  $10^{-7}$  nT as opposed to the theoretical value of 0.001 nT and the operationally realized value of 0.01 nT in current fleet systems (Chilton) [Ref. 20]. Field changes on the order of  $10^{-5}$  nT have actually been measured. These SQUIDS are used in both superconducting magnetometers, and superconducting gradiometers.

A Josephson junction consists of a thin layer of insulator between two superconductors. This junction has the property that a current can flow across it without developing a voltage up to some maximum current. A voltage is developed for all current values greater than the maximum current.

In a superconducting magnetometer a superconducting ring containing a pair of Josephson junctions is used to measure the amount of magnetic flux penetrating the loop. This system is a vector magnetometer which measures variations in only one field component.

By using more than one superconducting loop, it is possible to measure magnetic field gradients, and construct a superconducting gradiometer. This is also a component, or vector, sensor.

These instruments have been used extensively in the laboratory and to some extent in geophysical work, but considerable engineering problems remain to be solved before operational Navy use can be contemplated.

## C. MAD SIGNAL AND BANDPASS

### 1. Source of the Signal

The MAD signal results from moving a magnetometer through the magnetic field of a submarine, which can be approximated by the field of a magnetic dipole. Figure 2.4 depicts the formation of a submarine caused anomaly.

The magnetic moment of an object in the earth's magnetic field can be due to permanent magnetization, magnetization induced by the earth's magnetic field, or a combination of both. In the case of a submarine hull both causes are present with a small amount of permanent magnetization produced by hull stress in metal components during construction and stress caused by submarine diving and surfacing. This, however, is a minor contribution. The most important constituent of the submarine magnetic moment is due to magnetization induced in the hull by the presence of the geomagnetic field. [Ref. 21]

The induced field of a submarine depends on the effective permeability along the vertical, athwartships, and longitudinal axes of the submarine. This information, taken together with the strength and dip angle of the geomagnetic field, yields a fairly precise calculation of the magnetic anomaly produced by the submarine. 'Deperming'

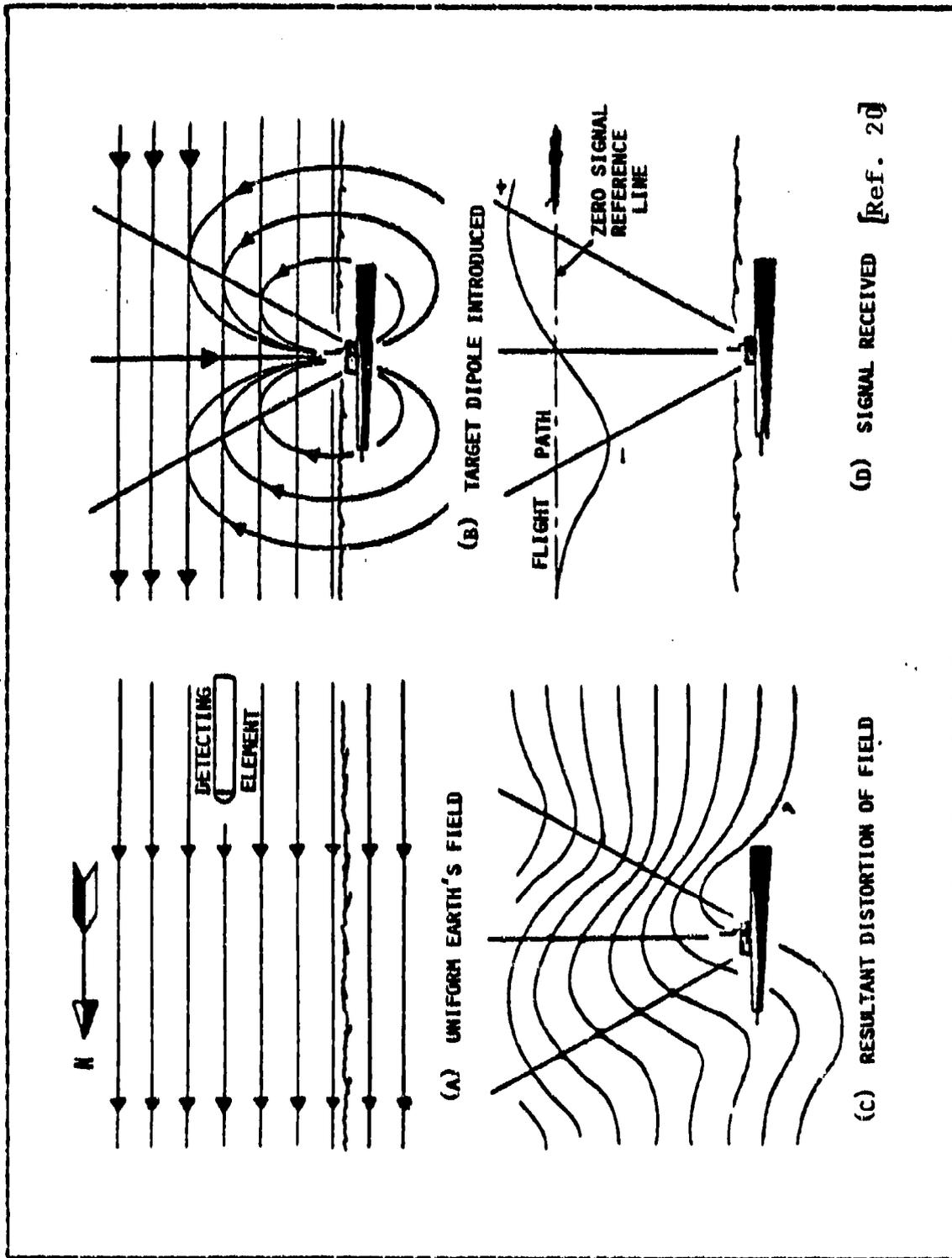


Figure 2.4 Formation of Submarine Caused Magnetic Anomaly (Simplified View Near Equator)

of the submarine hull cancels out the permanent hull magnetization leaving the induced magnetization as the principal signal source. [Ref. 21]

The MAD signal is approximated as the projection of the submarine dipole field onto the geomagnetic field vector. This approximation is good because the magnitude of the dipole field is very much smaller than the magnitude of the geomagnetic field (a few nanoTeslas as opposed to approximately 50000 nT). Therefore, whenever the dipole field is perpendicular to the earth's field a region of zero signal will result. The signal recorded by the AN/ASQ-81 or other MAD equipment is a mapping along the aircraft's flight path of this signal. Figure 2.5 qualitatively describes some aspects of the MAD signal.

## 2. Anderson Functions

The submarine anomaly signal shape is a function of the dip angle of the geomagnetic field, the magnetic heading of the aircraft, magnetic heading of the submarine dipole, and the lateral range between the aircraft and the submarine. These factors determine the 'A' coefficients for the Anderson functions below.

In 1949, J. E. Anderson of the Naval Air Development Center determined that the MAD signal, obtained along any course, consisted of a linear combination of three basic components. Different shaped signals could be obtained by changing the proportional contribution of these basic components. The mathematical representation of these components are now referred to as the Anderson functions. Using the dimensionless parameter 'b' (defined as the distance traveled along the aircraft track divided by the slant range at closest point of approach (CPA),  $b=0$  at CPA) the anomaly can be represented by

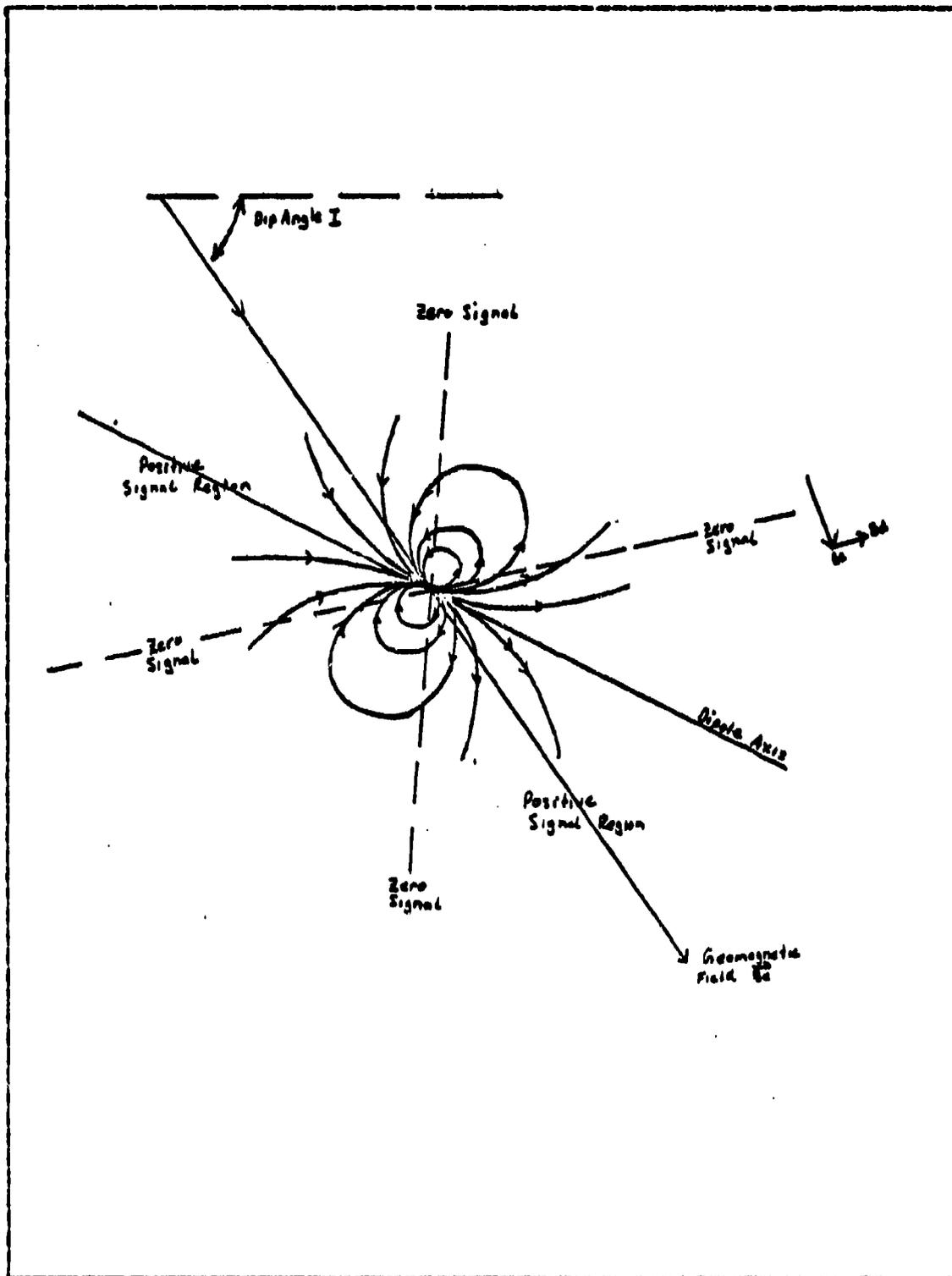


Figure 2.5 Qualitative Aspects of MAD Signal

$$B_s = \frac{M}{z^3} \frac{A_2 b^2 + A_1 b + A_0}{(b^2 + 1)^{5/2}} \quad (2-1a)$$

$$\text{or } B_s = \frac{M}{z^3} [A_0 f_0(b) + A_1 f_1(b) + A_2 f_2(b)] \quad (2-1b)$$

where  $M$  = magnetic moment of target dipole  
 $z$  = lateral range between target and plane at CPA

$$f_0(b) = \frac{1}{(b^2 + 1)^{5/2}}$$

$$f_1(b) = \frac{b}{(b^2 + 1)^{5/2}}$$

$$f_2(b) = \frac{b^2}{(b^2 + 1)^{5/2}}$$

The strength of the detected signal is seen to fall off as the cube of the distance between the aircraft and the target.

By analyzing the functions and their coefficients, it can be shown that the optimum orientation for maximum anomaly signal detection is when the aircraft, target dipole, and geomagnetic field are lined up together as closely as possible. Specifically, this occurs when the submarine moment and the aircraft's track are oriented North-South. [Ref. 14, 21]

The Fourier transform of the component functions were taken to determine the frequency distribution of energy in the MAD signal. The anomaly signal components are shown in Figures 2.6 through 2.8 (Anderson) [Ref. 14]. The Fourier transforms of the signal components with a platform velocity of 150 knots are shown in Figure 2.9.

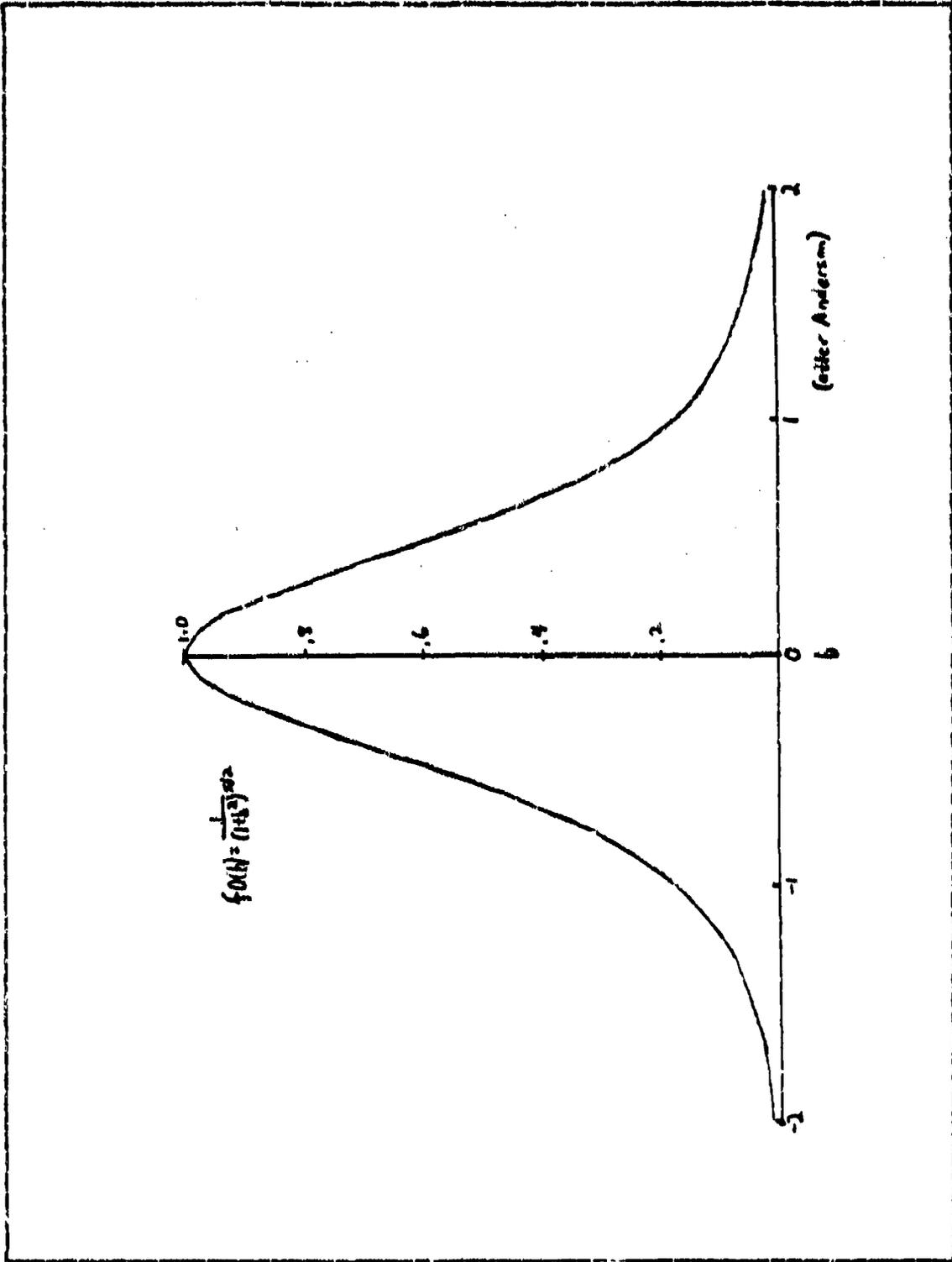


Figure 2.6 Basic MAD Component  $f_0(b)$

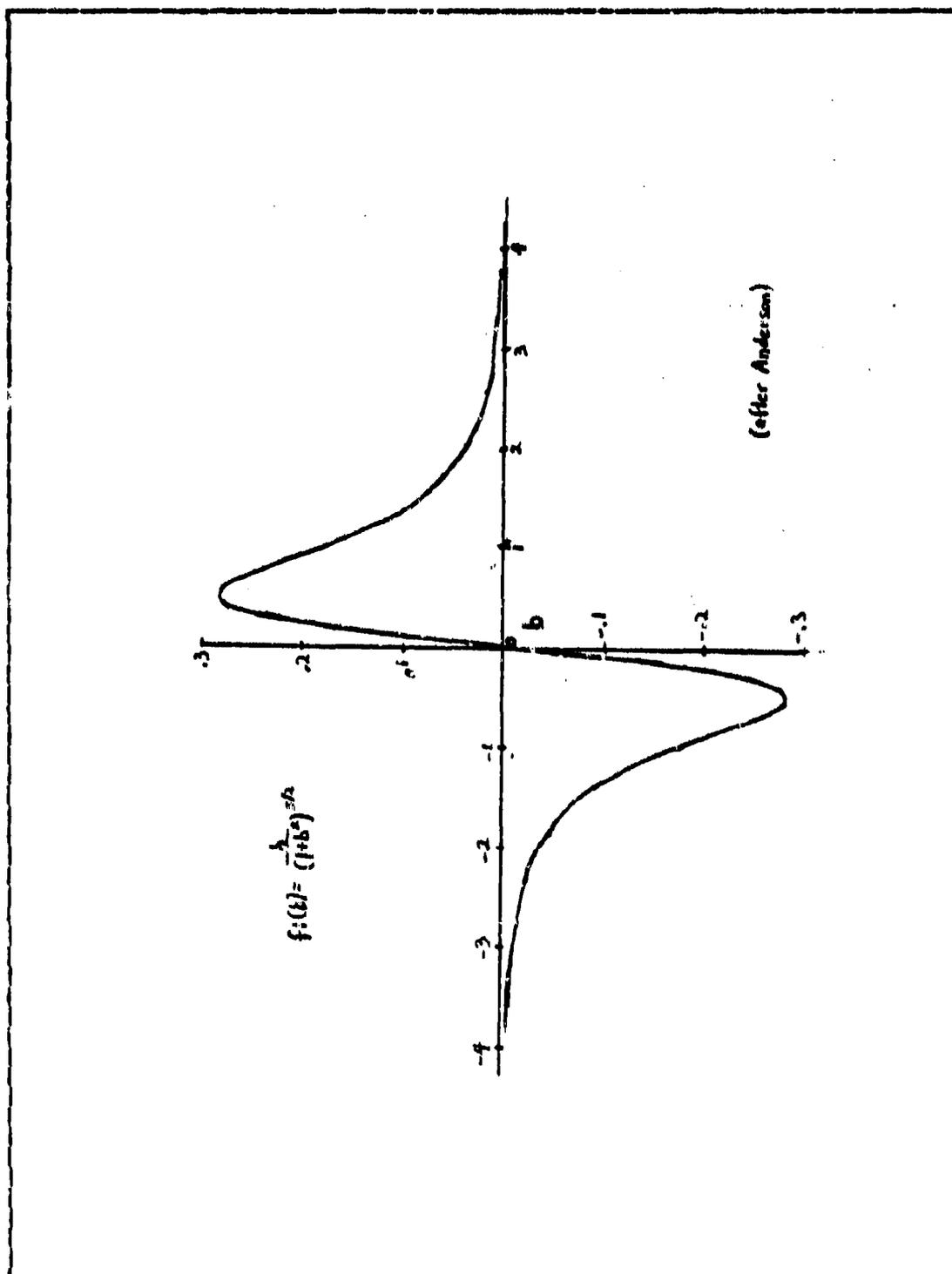


Figure 2.7 Basic MAD Component  $f_1(b)$

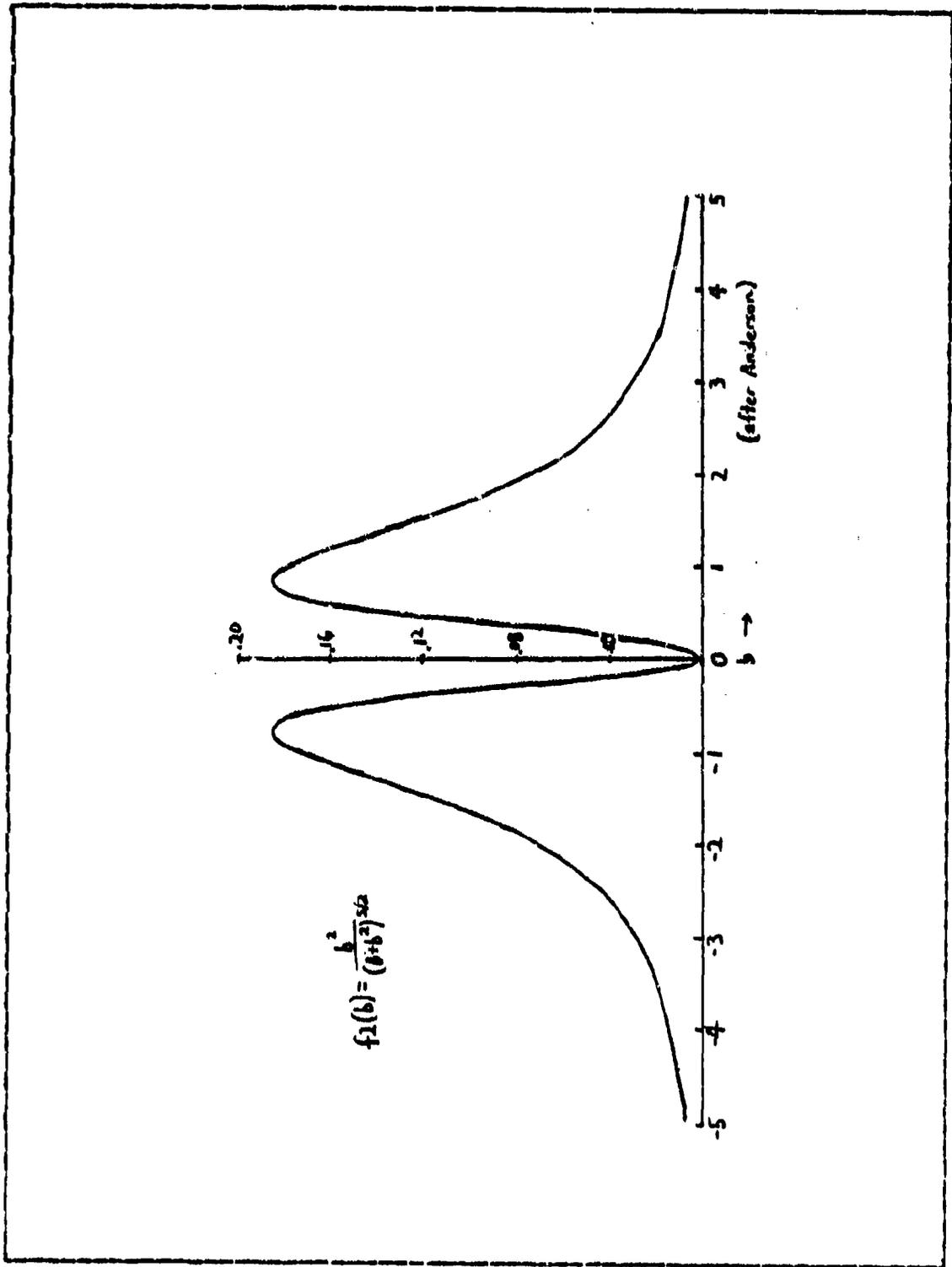


Figure 2.8 Basic MAD Component  $f_2(b)$

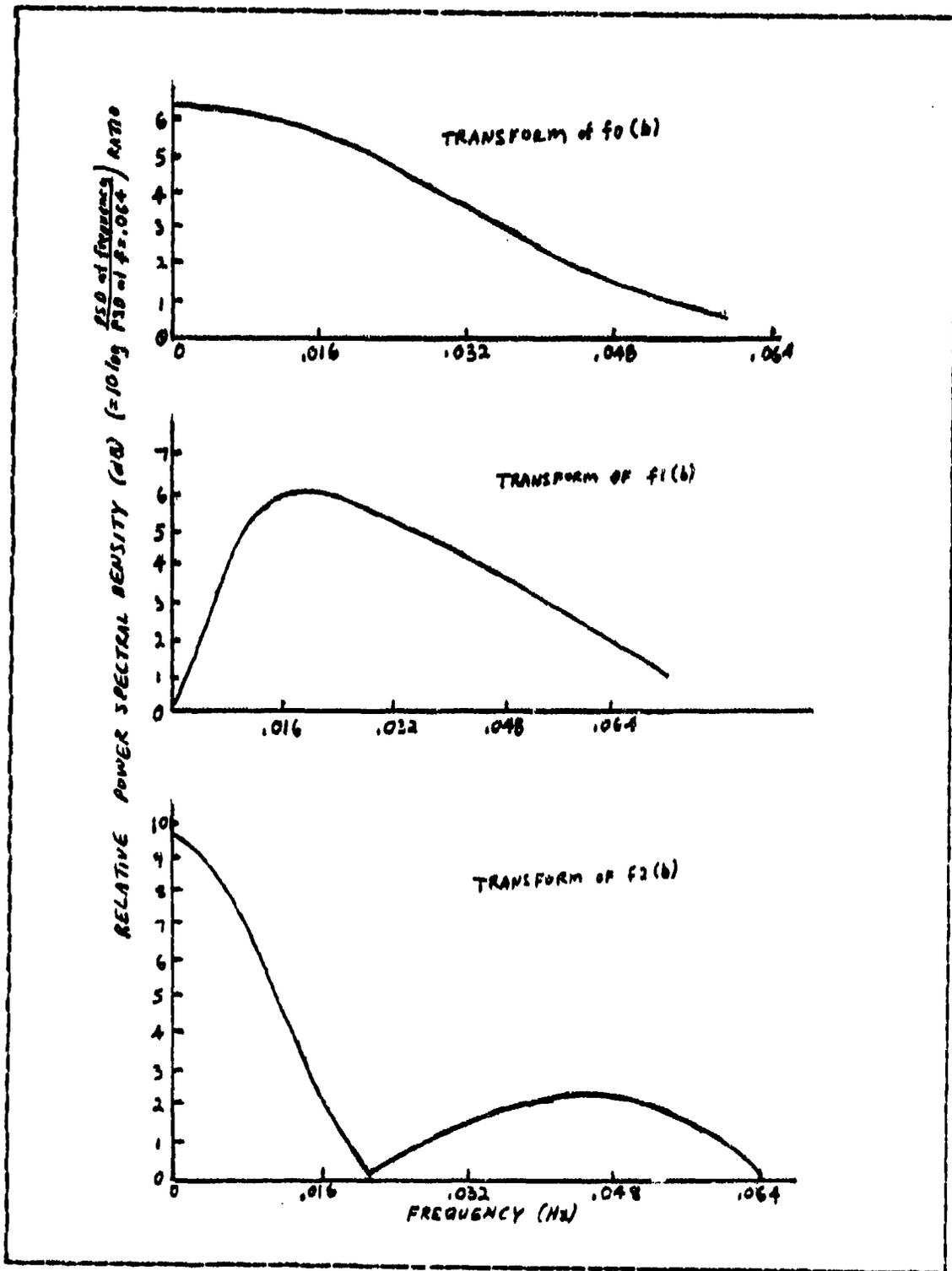


Figure 2.9 Frequency Spectrum (Fourier Transform) of MAD Signal Components

### 3. The MAD Filter Bandpass

The MAD signal, being a transient, is essentially a broadband (as opposed to a discrete frequency) signal. The frequency spectra of energy content shows how the signal is contained in a certain frequency range. In order to screen out unwanted noise, bandpass filtering is used in processing magnetic anomaly detections.

The major factors influencing the determination of an optimum MAD filter frequency are aircraft speed and the slant range from the aircraft to the target at CPA. Factors exerting a minor influence include dip angle, aircraft heading, and target dipole orientation.

The range in frequency variation is about an octave, with the higher values occurring when the passes are made parallel to the axis of the dipole and the lower when the pass is perpendicular to the dipole. Anderson empirically determined that the center frequency of the filter is given by

$$f = \frac{0.4 v}{Z} \quad (2-2)$$

where  $f$  = frequency in Hertz  
and  $v$  = aircraft velocity in ft/sec  
and  $Z$  = greatest anticipated CPA range in feet

The ASQ-81 Bandpass Filter has highpass settings of 0.04, 0.06, 0.08, and 0.1 Hz, and lowpass settings of 0.2, 0.4, 0.6, and 2.0 Hz (Orion Service Digest 26) [Ref. 22]. There are no recommended settings for normal operation since background noise varies. The filter characteristics for the 0.06 to 0.6 Hz settings are shown in Figure 2.10 (Orion Service Digest 28) [Ref. 23].

The figure is a representation of the adjustable band pass filter of the ASQ-81 by itself. Other parts of the ASQ-81 system add in an additional high pass filter which adds another 12 dB/octave roll off to the low frequency end in Figure 2.10. Thus Figure 2.10 would represent the characteristics of the entire ASQ-81 system if the roll off of 36 dB/octave is changed to 48 dB/octave. [Ref. 24]

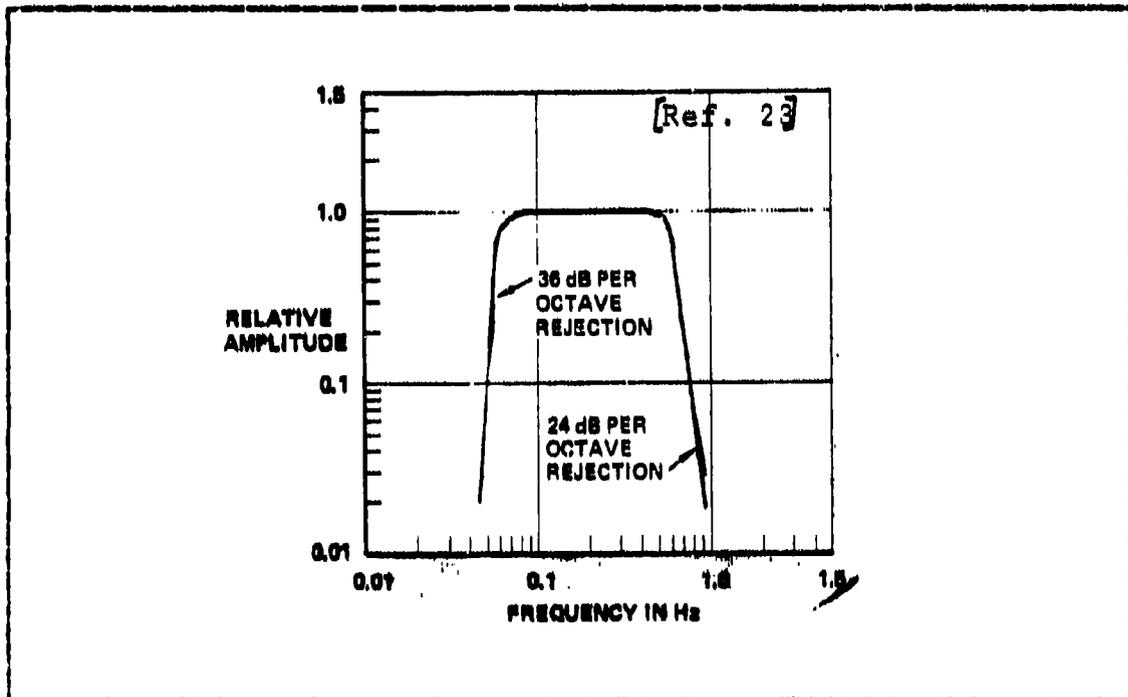


Figure 2.10 ASQ-81 Filter Characteristics: 0.06 - 0.6 Hz

Using equation 2-2, the center frequency for a MAD signal with a CPA slant range of 300 feet and aircraft velocity of 220 knots is 0.49 Hz. Using a lowpass filter setting above 0.6 Hz would not be very useful.

One reason the ASQ-81 filter extends up to 2.0 Hz instead of just 0.6 Hz is the following. The early Service Test Engineering Model (STEM) ASQ-81 that preceeded the

current ASQ-81 production model had two bandpass filters, one adjustable from .04 to .6 Hz, and the other fixed at .75 to 10 Hz. During tests, the .75 to 10 Hz channel proved to be helpful in monitoring the STEM ASQ-81 "system" noise level. During design reviews for the production ASQ-81, it was decided that it might be useful to retain the ability to monitor "system" noise. In order to do this easily, the .75 to 10 Hz band was dropped and a 2 Hz position was added to the adjustable filter. [Ref. 24]

The center frequency at 1200 feet and 160 knots is .09 Hz. Consequently, the highpass filter settings could affect slower, longer range MAD detections.

The choice of the highpass (lower end of bandpass) settings have to be made after considering the noise present at time of system operation.

### III. SOURCES OF MAD NOISE

#### A. INTRODUCTION

On a practical level MAD noise is defined as magnetic disturbances falling within the MAD passband (0.04 to 2.0 Hz, or since 0.6 Hz is the normal upper limit, 0.04 to 0.6 Hz) and having an amplitude greater than 0.01 nanoteslas (the system sensitivity of the AN/ASQ-81 MAD system).

MAD noise sources can be divided into the following categories:

- Equipment Noise
- Aircraft Platform Noise
- Aircraft Maneuver Noise
- Gradient Noise Due to Aircraft Motion Through the Geomagnetic Field
- Geologic Noise
- Noise from Wind Waves and Swells
- Geomagnetic Noise

#### B. SENSOR, PLATFORM AND MANUEVER NOISE

Sensor noise is the self-noise generated by the operation of the equipment itself. This can be partially due to the fact that the detector element is misaligned with respect to the geomagnetic field vector. Changes in lamp intensity, photodetector noise and noise in the electronic circuits can also contribute to sensor noise [Ref. 21]. The self-noise limitation of the ASQ-81 is 0.01 nT.

Platform noise is generated by components fixed to the aircraft in the vicinity of the sensor [Ref. 23]. Permanent, induced and eddy-current magnetic fields are

associated with the airframe. Permanent magnetic fields are due to aircraft structure or equipment having ferromagnetic parts. This field changes its orientation with respect to the geomagnetic field vector as the aircraft maneuvers, causing field fluctuations near the magnetometer.

Platform noise is also induced in aircraft ferromagnetic structures by the geomagnetic field. Similarly, eddy-currents are induced in aircraft skin, ribs, and frames, and these currents, in turn, cause additional magnetic fields. Thus, rapid aircraft maneuvers will induce changes in the magnetic field sensed by the magnetometer.

Platform noise in aircraft mounted sensors is countered by applying equal and opposite magnetic fields to the sensor in a process called compensation. Towed MAD systems are essentially free from this type of noise.

### C. GRADIENT NOISE

Gradient noise can be divided into turn noise and noise due to changes in altitude.

Turn noise is a problem when 'MAD trapping' or using MAD for tracking a target. The earth's magnetic field has a horizontal gradient (in this case the magnitude varying with latitude). As the aircraft moves in the direction of the gradient the field strength changes. The noise due to a MAD trapping or hunting circle is centered in frequency at the reciprocal of the time taken to complete one revolution. In the case of a two-minute circle, the noise would be centered at 0.00333 Hz, well below the filter used in the ASQ-81. The horizontal gradient noise due to flying a cloverleaf pattern, for the most part, also falls below the MAD passband. [Ref. 21]

Vertical gradient noise is due to changes in sensor altitude. An altitude gradient of up to 0.005 to 0.01 nT

per foot exists in the earth's main field. In areas of geological anomalies this gradient is even larger. Fast altitude or aircraft pitch changes can cause a magnetic field fluctuation of sufficient amplitude to be of concern. To avoid vertical gradient noise, altitude compensation equipment is used. [Ref. 23]

#### D. ENVIRONMENTAL NOISE

Magnetic noise from sources existing in the natural environment include geologic noise, temporal variation in the earth's magnetic field, and noise due to ocean waves and swells.

Geologic noise has its source in naturally occurring magnetic anomalies caused by magnetic material present in the earth's crust. When the sensor passes over geological anomalies, the relative motion causes a MAD-like signal to be recorded. Geologic noise is usually more pronounced in shallow water as the sensor is much closer to the source of the noise. Geological magnetic anomalies are often associated with such oceanographic features as seamounts and ocean ridges.

Sea water is a conducting medium which is transported by the physical motion of water waves in the presence of the geomagnetic field. This motion induces currents in the sea. These currents give rise to secondary magnetic fields, which add vectorially to the quasistatic, geomagnetic field (Weaver) [Ref. 25]. These fields can be detected at significant distances above the sea surface and fall off exponentially with altitude. Figure 3.1 is a plot for several surface wave periods of the induced magnetic field per meter amplitude of the surface wave. These induced fields can be a problem at the low altitudes where MAD is used.

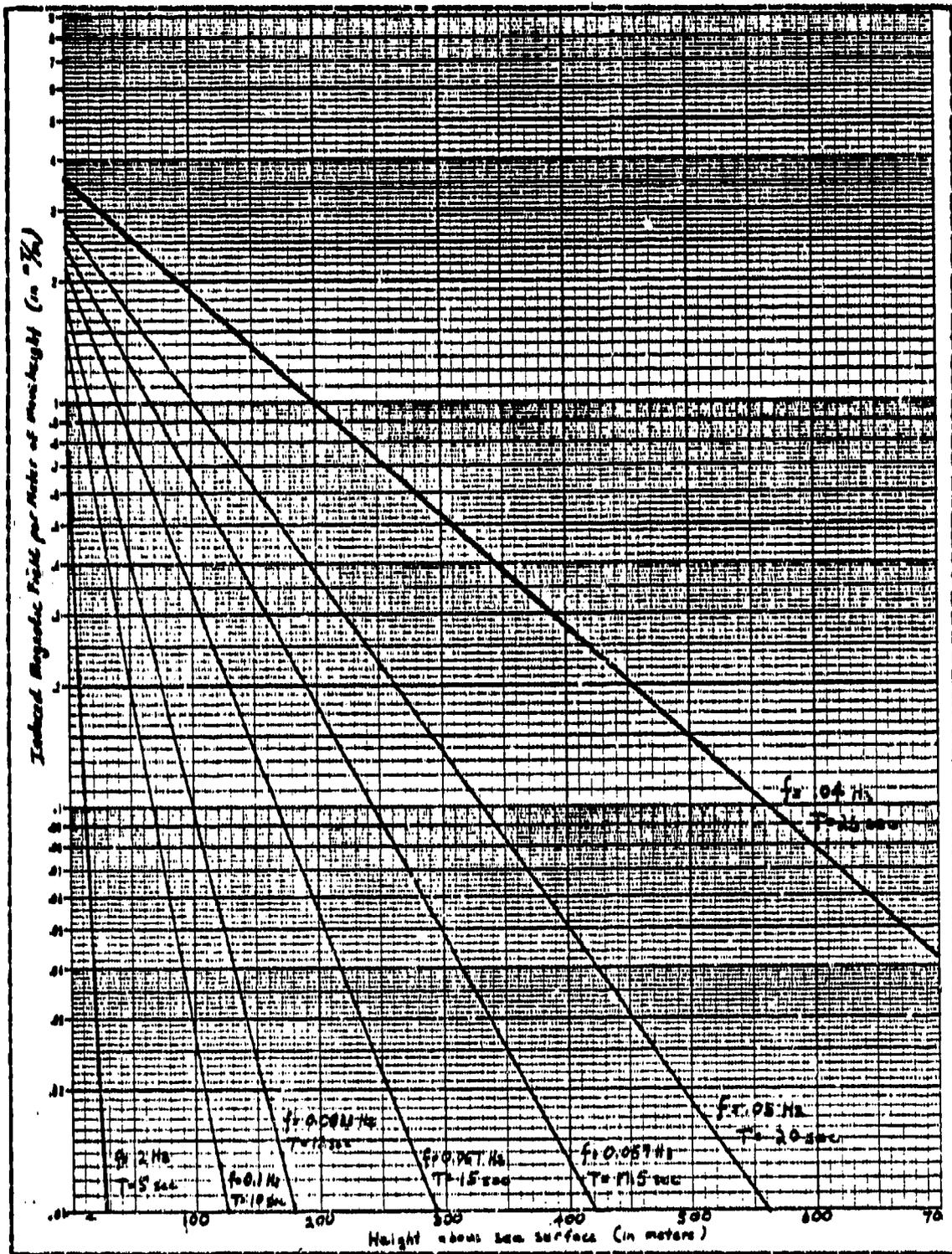


Figure 3.1 Induced Magnetic Field per Meter of Waveheight

## E. GEOMAGNETIC NOISE

Temporal variations in the earth's magnetic field with frequencies in the MAD bandpass and amplitudes greater than 0.01 nT have become known as geomagnetic noise in the MAD literature.

Quiet daily variations, such as the Sq and L variations, have periods sufficiently long to fall far below the MAD passband.

Geomagnetic storms have been discussed previously. Rapid fluctuations with high amplitude and falling within the MAD passband occur in connection with geomagnetic storms.

Geomagnetic micropulsations comprise the last category of geomagnetic noise to be discussed.

### 1. Geomagnetic Micropulsations

Geomagnetic micropulsations are rapid fluctuations of the earth's magnetic field with periods from 0.2 seconds to 10 minutes and amplitudes from about 0.1 nT to as high as a few tens of nT's. These fluctuations are caused by electromagnetic perturbations propagating in the magnetosphere as hydromagnetic waves (Nishida). [Ref. 26]

Micropulsations are classified by morphology, that is, by examining periods, amplitudes, times of occurrence and other observed characteristics. Micropulsations have been placed into two broad categories: irregular and continuous. The irregular pulsations are represented by the symbol Pi, and the continuous pulsations by Pc. Table II shows the breakdown by period of the various micropulsation classes.

It can be seen that for MAD geomagnetic noise the lower frequency Pc1, Pc2, Pc3, and Pi1 pulsations are of interest.

TABLE II

## Geomagnetic Micropulsation Classes

NOTATION	PERIOD (sec)	AVERAGE AMPLITUDE (nT)
Pc1	0.2 - 5.0	0.05 - 0.1
Pc2	5 - 10	0.1 - 1.0
Pc3	10 - 45	0.1 - 1.0
Pc4	45 - 150	0.1 - 1.0
Pc5	150 - 600	0.1 - 10
Pi1	1 - 40	0.05 - 0.1
Pi2	40 - 150	1 - 5

(after Jacobs)

Pc1 pulsations are regular sinusoidal oscillations with periods normally falling in the 0.3 to 4 second (0.25 to 3.33 Hz frequency) range. Pc1's may start as separate bursts and gradually develop into a series of pulsations which could last for hours. They may also occur as consecutive groups of pulsations with sharply varying frequencies. The average amplitude of Pc1's is 0.05 to 0.1 nT and they tend to have a single well defined frequency. Pc1 pulsations with frequency less than 0.5 Hz are more common at high latitudes than at mid latitudes. These pulsations occur in the daytime in the auroral zone, and at night and early morning hours in lower latitudes. Pc1's are characteristic of the quiet and weakly disturbed states of the geomagnetic field and show an increase in activity one to two hours before, and four to seven days after, a magnetic storm. [Ref. 6]

Pc2 and Pc3 pulsations are grouped together in most characteristics. The amplitudes of the oscillations are usually under 0.5 nT and the typical frequency range is 0.03 to 0.2 Hz. These pulsations are normally found during the day with their activity reaching a maximum around noon. Pc3 pulsations show a seasonal variation with a minimum of activity occurring during winter.

Pc4 and Pc5 are large amplitude fluctuations, but fall below the frequency band of interest for MAD.

Pi1 pulsations have an irregular form with an average amplitude of 0.01 to 0.1 nT and a frequency mainly in the 0.10 to 0.17 Hz range. Spectral analysis of these pulsations show a broad band of frequencies. Pi1 amplitudes have maximum values in the auroral zones, with the intensity of the pulsations decreasing with decreasing latitude. Pi1's are normally observed in the late night and early morning hours, and show an increase in activity with increased geomagnetic field disturbance. [Ref. 6]

Geomagnetic micropulsations can be observed anywhere on the globe, at various times of day and year, and in both quiet and disturbed geomagnetic field conditions.

## IV. METHODS OF EVALUATING GEOMAGNETIC ACTIVITY

### A. INTRODUCTION

Previous chapters have defined geomagnetic activity as it applies to Magnetic Anomaly Detection. In this chapter, methods of evaluating that activity will be examined, including methods currently in use by the fleet.

### B. GEOMAGNETIC INDICES

A geomagnetic index is simply a measure used to quantify and describe time variations of the earth's magnetic field resulting from solar-terrestrial relationships. These indices are commonly used to express the intensity and depict the character of geomagnetic activity throughout the day.

For the most part, geomagnetic indices developed as range indices, measuring the difference between the high and low values for different field components measured during the day by magnetic observatories (Lincoln) [Ref. 27]. Most current indices are of the range type, but other indices have been developed which are more subjective or qualitative in nature.

Geomagnetic indices are designated by a letter code such as: C, Ci, Cp, C9, Q, R, W measure, Dst, K, Ks, Kp, ak, Ak, ap, and Ap. There are additional indices in use.

The C, Ci, Cp, and C9 indices are daily magnetic field character figures. The C index is the daily character figure for a single observatory. In this scale, C=0 indicates a quiet day, C=1 a moderately disturbed day, and C=2 a heavily disturbed day. The daily international character

figure,  $C_i$ , is the arithmetic mean of the C indices reported by participating observatories around the world.  $C_p$ , the daily planetary character figure, is similar to  $C_i$ , except that it is derived from the values of  $K_p$  and  $a_p$ .  $C_9$  is a contracted scale for  $C_i$  or  $C_p$  with single digit values running from 0 to 9 (Bartals). [Ref. 28].

The Q and R indices are quarter-hourly and hourly range indices respectively, taken at high latitude stations only.

The W measure is an index of the equatorial electrojet. Dst is a measure of ring current effect. They are both amplitude indices. [Ref. 27]

### 1. K, $a$ , and A Indices

The K,  $K_s$ ,  $K_p$ ,  $a_k$ ,  $A_k$ ,  $a_p$ , and  $A_p$  indices comprise a group of related 3-hour range indices. The K index is a single station code using a quasi-logarithmic scale from 0 to 9 to measure geomagnetic activity. The value of K is determined by first determining the difference between the lowest and highest deviations from the regular daily variation ( $S_q$ ) during a 3-hour period. This range (in nT) is converted to the K scale based on the historical activity ranges at the particular observatory involved [Ref. 27]. The conversion for the Fredericksburg, Virginia observatory is given in Table III. This conversion can also be applied to the USAF/NOAA observatory in Boulder, Colorado [Ref. 29].

The  $K_s$  index is a standardized K index which is freed from local variations and is then used to determine the planetary 3-hour index,  $K_p$ .

The equivalent three-hour-range,  $a_k$ , is a conversion of the K index as shown in Table IV. In order to determine the units of  $a_k$  for a particular observatory, divide the lower range limit of  $K=9$  by 250. Thus for Fredericksburg and Boulder,  $a_k$  is in 2-nT units.

TABLE III

Conversion from Range to K for Fredericksburg, Va.

K	Range (nT)	
0	50	5
1	100	10
2	200	20
3	400	40
4	700	70
5	1200	120
6	2000	200
7	3300	330
8	5000	500
9	> 5000	> 500

(after Lincoln)

TABLE IV

Equivalent Range ak for Given K

K	0	1	2	3	4	5	6	7	8	9
ak	0	3	7	15	27	48	80	140	240	400

(after Lincoln)

Ak is the equivalent daily amplitude and is the average of the eight daily ak values at a particular observatory. This index is promulgated using the name of the observatory, the Ak index for Fredericksburg is known as the A-Fredericksburg or A-Fred index.

The equivalent planetary amplitude, ap, is determined from the Kp index in a fashion similar to that of determining ak from the K indices. The eight ap values for a given day can then be averaged into the daily equivalent planetary amplitude Ap. These two indices are given in 2-nT units.

## C. GEOMAGNETIC INDICES IN FLEET USE

### 1. Current Usage

Geomagnetic indices of interest in connection with MAD operations are the K, Ak, and Ap indices.

Fleet operators utilize the "alpha Index" for predicting geomagnetic activity over the entire world [Ref. 30]. This index is promulgated by the Fleet Numerical Oceanographic Center, Monterey, California, in the environmental briefings received by aircrew personnel. This index is the Ap index as sent out from the Space Environmental Services Center, Boulder, Colorado in the Joint USAF/NOAA Report of Solar and Geophysical Activity [Ref. 31].

The Boulder K index is available to interested parties by telephone recording and in the WWV and WWVH radio broadcasts and is therefore available to fleet users [Ref. 32].

The Ak index for the Fredericksburg, Virginia observatory has been used in studies of geomagnetic activity as applied to Magnetic Anomaly Detection [Ref. 33].

### 2. Theoretical Applicability of A and K Indices to MAD

With the widest useful filter settings, the MAD bandpass ranges from 0.04 to 0.6 Hz (1.7 to 25 seconds in period). As such, in order for a geomagnetic index to be directly applicable for MAD use, it should be sensitive to that frequency range.

The K indices, and the K-derived A indices, are not especially sensitive to the MAD range. Mayaud [Ref. 34] indicates that these indices are mainly sensitive to fluctuations whose periods are much longer than the lower end of the frequency range analyzed, that is, a frequency corresponding to a period of 45 minutes (0.0004 Hz).

One reason for this lack of sensitivity for MAD bandpass geomagnetic noise is that the amplitude of geomagnetic fluctuations varies inversely with frequency, so that the amplitude of the fluctuation increases as the frequency decreases. It can therefore be seen that the fluctuations with periods of an hour or greater largely determine the variation range used to calculate the K index. The activity driving the K and A indices, will, because of band pass filtering, not even be observed by the MAD system, and the activity of interest to MAD might not influence the K or A indices at all. It can be concluded that there is no direct physical link between MAD geomagnetic noise and either the K or A indices. [Ref. 35]

### 3. Experimental Correlation of A and K Indices with MAD Band Noise

#### a. ASQ-10A Study

Brennan and Smits [Ref. 33] found that geomagnetic micropulsation activity was recorded at their ASQ-10A MAD magnetometer site in Maryland, when the A-Fredericksburg index was greater than 25. This occurred everytime they were recording data with A-Fred greater than 25. It is important to note that they observed additional activity during some periods when A-Fred was less than 25.

The Brennan and Smits study tends to validate the use of the A indices as at least qualitative indications of geomagnetic noise in the MAD bandpass. Their study, however, was specific to the ASQ-10A magnetometer system which has a sensitivity of 0.1 nT as opposed to the 0.01 nT sensitivity of the ASQ-81 system. The effect of the ASQ-10 sensitivity is to filter out most Pc1 pulsations. As Pc1 pulsations do not correlate well with the A and K indices, the filtering out of these pulsations would tend to increase

the reliability of the A-Fred or Ap indices as measures of MAD geomagnetic noise. Mason [Ref. 36] stated that the "occurrence of Pc1 is well known to be associated with low Kp values." It should be noted that an operational drawback of the filtering out of Pc1 pulsations by the ASQ-10A is that the system also filters out valid signals of less than 0.1 nT amplitude.

While information has been presented that suggests the the A-Fred index can be useful for MAD geomagnetic noise evaluation for the AN/ASQ-10A system, sufficient data was not presented to draw conclusions for index usage with the AN/ASQ-81 system.

#### b. ASQ-81 Study

For two weeks in April 1976, Naval Air Development Center personnel operated a geomagnetic observatory at the Atlantic Undersea Test and Evaluation Center in the Bahamas. The primary magnetometer used for this observatory was the ASQ-81 magnetometer. One purpose of this observatory was to compare the K index (as determined by the San Juan Observatory) with geomagnetic activity in the MAD band pass. The conclusion of this study was that the K-San Juan index did not correlate with geomagnetic noise in the MAD band pass. [Ref. 37]

We have undertaken a correlation analysis using the Fisher Z transformation [Ref. 38] of the NADC data furnished to us by Ochadlick. The data used is listed in Tables V and VI.

TABLE V

Observed AUTEC Data and Indices, April 11-18, 1976.

Date	Fi	K-Sa	Juan	A	Am	HT
11-11	100000	000000	000000	000000	000000	000000
11-12	100000	000000	000000	000000	000000	000000
11-13	100000	000000	000000	000000	000000	000000
11-14	100000	000000	000000	000000	000000	000000
11-15	100000	000000	000000	000000	000000	000000
11-16	100000	000000	000000	000000	000000	000000
11-17	100000	000000	000000	000000	000000	000000
11-18	100000	000000	000000	000000	000000	000000

(a) rter Ocha



being from 0.14 to to 0.52. Sample size was 86. This indicates that there is at most a weak correlation between the observed data and the K-San Juan index. A much greater correlation would be required for the K index to be of any significant value for MAD operational use.

The coefficient of correlation has values from -1.0 to 1.0. A value of -1.0 or 1.0 indicates perfect negative or positive correlation, respectively. A value of zero signifies no correlation at all.

A similar correlation analysis against the Ap index was conducted. The sample correlation coefficient was 0.51 with the .95 confidence interval for the actual coefficient of correlation being from 0.32 to 0.65. Again, this signifies that only a weak correlation was observed.

The weak correlation between observed geomagnetic activity and the K-San Juan index suggests that this index would not be very useful in describing MAD geomagnetic noise, as this K index reflected activity similar to that in the band of interest only about one-third of the time. The stronger correlation of the Ap index indicates that it reflected activity similar to MAD geomagnetic noise about one-half of the time. This is still not a very good indication of what is going on in the MAD band pass.

#### c. Power Spectral Density Evaluation

As part of ongoing research at the Naval Postgraduate School, geomagnetic activity data in the range of the MAD band pass has been collected and analyzed in the form of power spectral density (PSD) curves.

A measure of MAD band activity has been developed by integrating under the PSD curve and using compromise band pass limits of 0.05 and 1.0 Hz. This type of index is discussed in depth later.

Two sets of data were analyzed. One set was taken using an induction coil to measure the fluctuations in one direction during May, 1980 [Ref. 39] and August to October, 1981 [Ref. 40]. The other data set was collected using a Cesium vapor total field magnetometer during the July to October, 1980 period [Ref. 41].

Correlation analysis of the single-coil system data yielded a sample correlation coefficient of 0.401 for the K-Fredericksburg index, 0.180 for the A-Fredericksburg index and 0.155 for the Ap index. Sample size was 9. The .95 confidence intervals for the correlation coefficient for K-Fred, A-Fred, and Ap were -.36 to .85, -.55 to .76, and -.56 to .75, respectively. The data for this test is presented in Table VII.

TABLE VII

Single Coil RMS Noise Data and Indices

Date	Time (Z)	RMS Noise Amplitude (nF)	K-Fred	A-Fred	Ap
May 1	1800	0.10			1.45
Aug 1	2040	0.05			1.60
Aug 1	0234	0.12			1.60
Aug 1	0745	0.13			1.60
Aug 1	1318	0.05			1.60
Oct 1	1830	0.05			1.77
Oct 1	0150	0.12			1.77
Oct 1	0630	0.03			1.77
Oct 1	1530	0.02			1.77

The Cesium vapor magnetometer data yielded sample coefficients of correlation of 0.552, 0.374, and 0.444 for the K-Fred, A-Fred, and Ap indices, respectively. The sample size was 14. The .95 confidence intervals were for K-Fred from .03 to .84, A-Fred from -.20 to .76, and for Ap from -.12 to .79. The Cs vapor magnetometer data is presented in Table VIII.

TABLE VIII

Cs Vapor RMS Noise Data and Indices, Jul-Oct, 1980

Time	RMS Noise Amplitude (nT)	K-Fred	A-Fred	Ap
0000	0.04	1	3	1
0005	0.04	1	3	1
0010	0.04	1	3	1
0015	0.04	1	3	1
0020	0.04	1	3	1
0025	0.04	1	3	1
0030	0.04	1	3	1
0035	0.04	1	3	1
0040	0.04	1	3	1
0045	0.04	1	3	1
0050	0.04	1	3	1
0055	0.04	1	3	1
0100	0.04	1	3	1
0105	0.04	1	3	1
0110	0.04	1	3	1
0115	0.04	1	3	1
0120	0.04	1	3	1
0125	0.04	1	3	1
0130	0.04	1	3	1
0135	0.04	1	3	1
0140	0.04	1	3	1
0145	0.04	1	3	1
0150	0.04	1	3	1
0155	0.04	1	3	1
0200	0.04	1	3	1
0205	0.04	1	3	1
0210	0.04	1	3	1
0215	0.04	1	3	1
0220	0.04	1	3	1
0225	0.04	1	3	1
0230	0.04	1	3	1
0235	0.04	1	3	1
0240	0.04	1	3	1
0245	0.04	1	3	1
0250	0.04	1	3	1
0255	0.04	1	3	1
0300	0.04	1	3	1
0305	0.04	1	3	1
0310	0.04	1	3	1
0315	0.04	1	3	1
0320	0.04	1	3	1
0325	0.04	1	3	1
0330	0.04	1	3	1
0335	0.04	1	3	1
0340	0.04	1	3	1
0345	0.04	1	3	1
0350	0.04	1	3	1
0355	0.04	1	3	1
0400	0.04	1	3	1
0405	0.04	1	3	1
0410	0.04	1	3	1
0415	0.04	1	3	1
0420	0.04	1	3	1
0425	0.04	1	3	1
0430	0.04	1	3	1
0435	0.04	1	3	1
0440	0.04	1	3	1
0445	0.04	1	3	1
0450	0.04	1	3	1
0455	0.04	1	3	1
0500	0.04	1	3	1
0505	0.04	1	3	1
0510	0.04	1	3	1
0515	0.04	1	3	1
0520	0.04	1	3	1
0525	0.04	1	3	1
0530	0.04	1	3	1
0535	0.04	1	3	1
0540	0.04	1	3	1
0545	0.04	1	3	1
0550	0.04	1	3	1
0555	0.04	1	3	1
0600	0.04	1	3	1
0605	0.04	1	3	1
0610	0.04	1	3	1
0615	0.04	1	3	1
0620	0.04	1	3	1
0625	0.04	1	3	1
0630	0.04	1	3	1
0635	0.04	1	3	1
0640	0.04	1	3	1
0645	0.04	1	3	1
0650	0.04	1	3	1
0655	0.04	1	3	1
0700	0.04	1	3	1
0705	0.04	1	3	1
0710	0.04	1	3	1
0715	0.04	1	3	1
0720	0.04	1	3	1
0725	0.04	1	3	1
0730	0.04	1	3	1
0735	0.04	1	3	1
0740	0.04	1	3	1
0745	0.04	1	3	1
0750	0.04	1	3	1
0755	0.04	1	3	1
0800	0.04	1	3	1
0805	0.04	1	3	1
0810	0.04	1	3	1
0815	0.04	1	3	1
0820	0.04	1	3	1
0825	0.04	1	3	1
0830	0.04	1	3	1
0835	0.04	1	3	1
0840	0.04	1	3	1
0845	0.04	1	3	1
0850	0.04	1	3	1
0855	0.04	1	3	1
0900	0.04	1	3	1
0905	0.04	1	3	1
0910	0.04	1	3	1
0915	0.04	1	3	1
0920	0.04	1	3	1
0925	0.04	1	3	1
0930	0.04	1	3	1
0935	0.04	1	3	1
0940	0.04	1	3	1
0945	0.04	1	3	1
0950	0.04	1	3	1
0955	0.04	1	3	1
1000	0.04	1	3	1
1005	0.04	1	3	1
1010	0.04	1	3	1
1015	0.04	1	3	1
1020	0.04	1	3	1
1025	0.04	1	3	1
1030	0.04	1	3	1
1035	0.04	1	3	1
1040	0.04	1	3	1
1045	0.04	1	3	1
1050	0.04	1	3	1
1055	0.04	1	3	1
1100	0.04	1	3	1
1105	0.04	1	3	1
1110	0.04	1	3	1
1115	0.04	1	3	1
1120	0.04	1	3	1
1125	0.04	1	3	1
1130	0.04	1	3	1
1135	0.04	1	3	1
1140	0.04	1	3	1
1145	0.04	1	3	1
1150	0.04	1	3	1
1155	0.04	1	3	1
1200	0.04	1	3	1
1205	0.04	1	3	1
1210	0.04	1	3	1
1215	0.04	1	3	1
1220	0.04	1	3	1
1225	0.04	1	3	1
1230	0.04	1	3	1
1235	0.04	1	3	1
1240	0.04	1	3	1
1245	0.04	1	3	1
1250	0.04	1	3	1
1255	0.04	1	3	1

Although the sample sizes used were too small to draw any meaningful conclusions, there is little evidence to suggest that any of the K-Fred, A-Fred, or Ap indices is a very accurate measure of geomagnetic noise in the ASQ-81 MAD band pass.

d. Correlation Conclusions

Although some weak correlation does exist between the K indices, the A-Fred index, the Ap index and geomagnetic noise in the MAD band pass, this correlation is incidental and indirect, being the result of a correlation between the activity in the MAD band and in the lower frequency activity that influences the K and A indices. These indices are not directly influenced by activity in the MAD band. The correlation that does exist does not appear to be sufficiently high to enable these indices to yield accurate indications of the actual MAD band activity. The use of these indices for anything except the roughest qualitative estimation of activity in the MAD band pass is not recommended.

#### D. PROPOSED GEOMAGNETIC INDICES FOR MAD

Overall geomagnetic activity is analyzed in both the time and frequency domains. Geomagnetic noise indices for the MAD band pass could be developed in either of these domains.

The spatial coherence of MAD geomagnetic noise has not yet been adequately determined. This information would be necessary in order to determine the number and location of mini-observatories for an operational MAD noise index.

##### 1. Time Series Analysis

One way to develop an index of MAD geomagnetic activity would be to establish mini-observatories near bases from which MAD operations are conducted. These observatories would use ASQ-81 magnetometers or different magnetometers with ASQ-81 filter networks, and could in real time record the geomagnetic noise in the MAD band. A measure such as the maximum peak-to-peak (or possibly the average peak-to-peak) noise in a given time period could then be disseminated to flight crews operating in the area covered by that index. Obviously, the spatial coherence of MAD band activity is important in making such a system work. This type of mini-observatory has been suggested by References 31 and 35.

##### 2. Frequency Domain Index

Present fleet procedures examine MAD noise such as system and maneuver noise in terms of the amplitude of the fluctuation [Ref. 30]. An index of geomagnetic noise in the MAD band pass would therefore be of greatest usefulness to the fleet operator if it were in units of the amplitude of the signal as seen by the MAD equipment.

The method proposed to derive a MAD noise index in the frequency domain begins with obtaining the power spectral density of the activity in the MAD band by Fourier analysis of the time series data input from the magnetometer.

By intergrating under the PSD curve over the limits of the MAD bandpass, a value in units of amplitude<sup>2</sup> will result. Taking the square root of this value will yield an RMS amplitude. Equation 4-1 represents the derivation of this index.

$$\text{MAD Index} = \left( \int_1^u \text{PSD}(f) df \right)^{1/2} \quad (4-1)$$

where MAD Index is in nT (gammas)  
 $f$  = frequency (Hz)  
 $u$  = upper bandpass limit  
 $l$  = lower bandpass limit  
 and PSD( $f$ ) = power spectral density  
 (nT<sup>2</sup>/hz)

The characteristics of the MAD filter (the filter not being an ideal bandpass filter) could be applied prior to the integration. The integration itself could be done by either a point-by-point numerical integration or by first modelling the PSD curve by polynomial curve fitting and then integrating the polynomial over the range of the band pass. It is anticipated that this could be done in close to real time by a digital computer, possibly by a desk top computer such as the HP9845.

The type of sensor utilized could be the ASQ-81 magnetometer, other total field magnetometer, or possibly an orthogonal 3-coil system whose signals can be combined to yield the projection on the total field vector of the fluctuations. A single coil system oriented in the direction of the earth's magnetic field vector could also be used.

A three coil system which is used to yield an RMS amplitude is currently in research use at the Naval Postgraduate School.

### 3. Predictions of Geomagnetic Activity

The proposed indices discussed above are intended to be real time measures of the geomagnetic noise in the MAD bandpass. Whether or not such activity can be predicted ahead of time needs to be looked into.

While there is no model for the background component of geomagnetic noise, work has been done on estimating the future activity of micropulsations, notably by Fraser-Smith in the case of Pc1 pulsations [Ref. 42, 43]. By extending the prediction technique for Pc1 pulsations to the Pc2, Pc3, and Pi1 pulsations, the occurrence of geomagnetic micropulsation activity in the MAD band might be predicted. Combining this prediction with real-time solar flare information should give the capability to disseminate real-time and estimated future MAD index values to fleet users.

Previous studies have led to the conclusion that the currently used geomagnetic indices are not accurate measures of geomagnetic noise in the MAD band pass. Experimental equipment has been utilized, and computer software written in order to confirm this conclusion, and to develop a replacement means of evaluating MAD geomagnetic noise.

#### A. EQUIPMENT CONFIGURATION

Experimental equipment, acquired as part of the Naval Postgraduate School geomagnetics research program, has been utilized in the effort to develop a usable MAD index. This equipment is in use in other projects of the geomagnetics research group. The sensors and associated equipment are set up for remote site operation with system monitoring and data analysis located at the Naval Postgraduate School. Descriptions of the data collection system and data analysis system follow below.

##### 1. Data Collection System

The data acquisition system illustrated in Figure 5.1 reveals the following major components:

- coil antenna sensors (3)
- preamplifiers (3)
- signal conditioners (amplifiers) (3)
- pulse code modulation system (1)
- radio transmitter (1)
- radio receiver (1)
- instrumentation tape recorder (1)

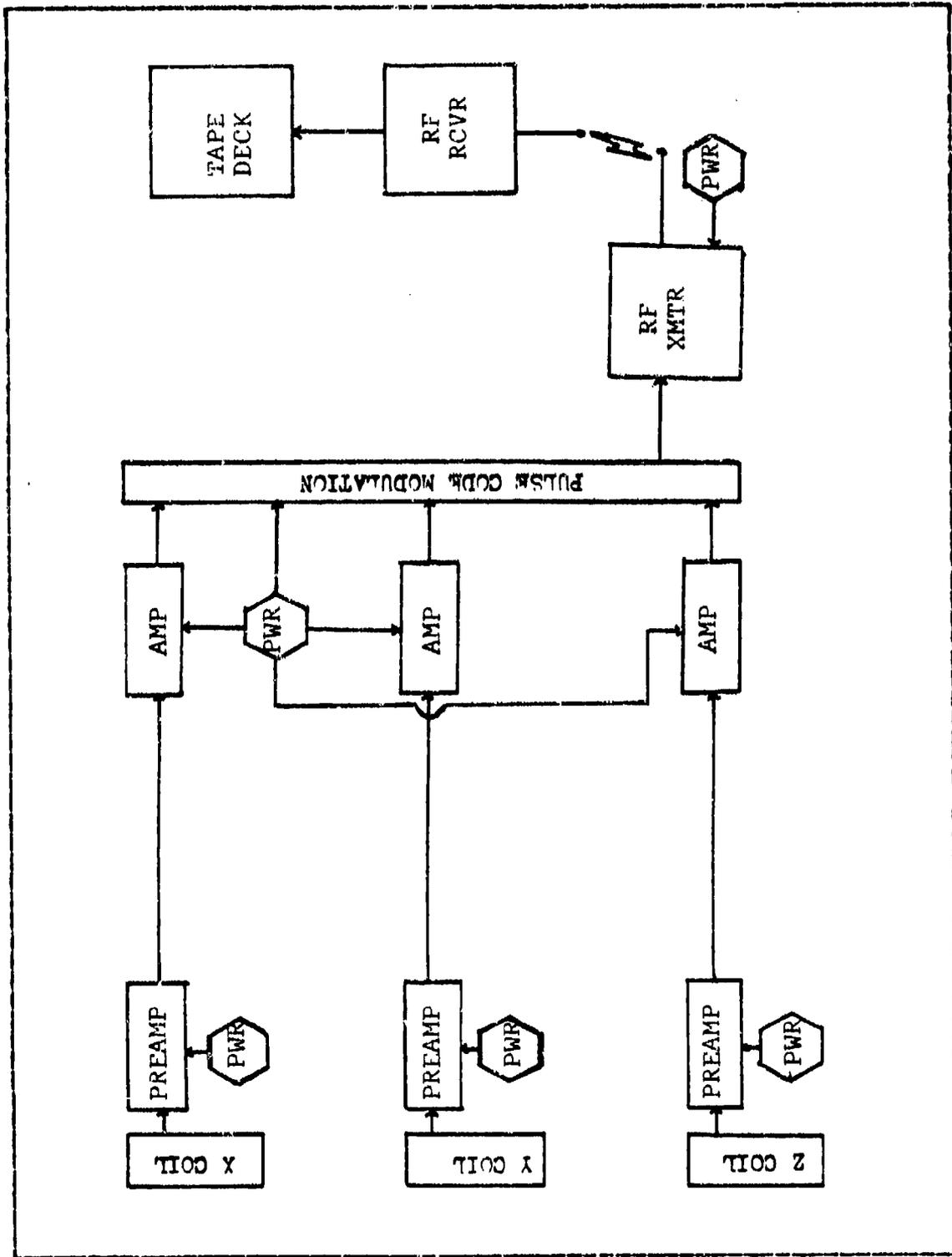


Figure 5.1 Data Collection System

### a. Coil Antenna Sensors

Three coil sensors are used in this system. Each sensor is a self-supporting, continuously wound, non-center-tapped coil antenna manufactured by Elma Engineering, Palo Alto, California, from about 5460 turns of 18 gauge copper magnet wire. The coils weigh approximately 50 kg each with dimensions as depicted in Figure 5.2. The dimensions of the sensor are constrained by the dimensions of the largest glass sphere that is commercially available. These spheres are used to enclose the coils during underwater experiments. The coil resistance is 120 ohms and its self-inductance is approximately 9.31 henries. The three coils are mounted orthogonally on a nonmagnetic frame (Figure 5.3).

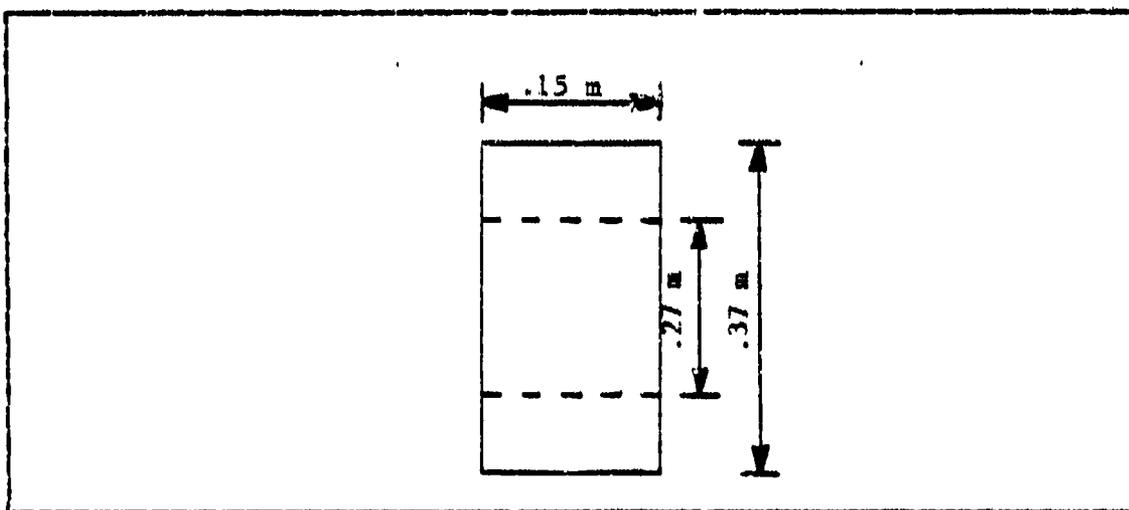


Figure 5.2 Sensor Dimensions

### b. Preamplifier

The preamplifier used was the model 13-10A low noise ELF amplifier manufactured by Dr. Alan Phillips of SRI

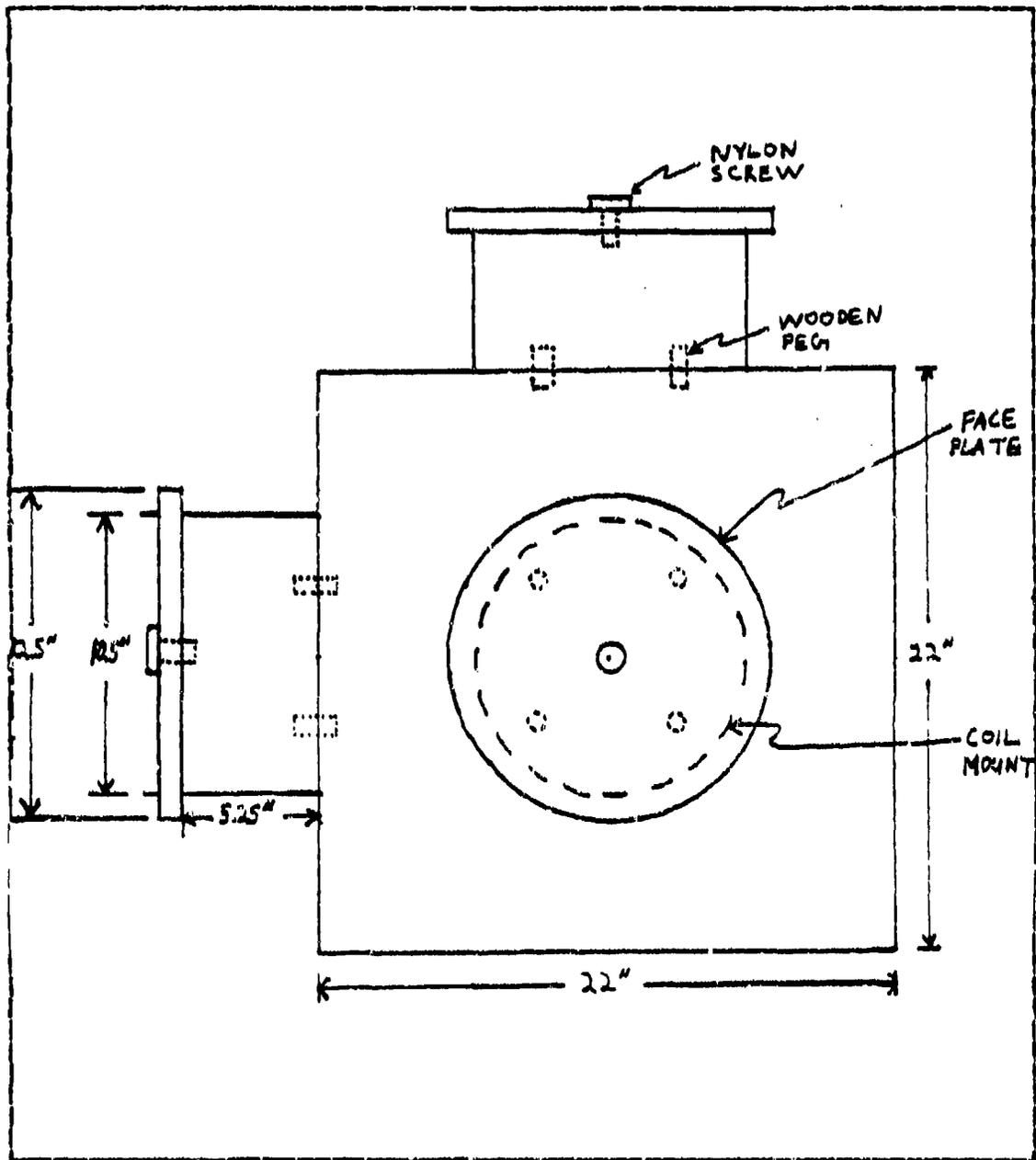


Figure 5.3 Sensor Mounting Block

International. The final stage of the amplifier contains an active low-pass filter which provides a sharp cutoff for frequencies above 20 Hz. The overall preamplifier gain for inputs of less than 2.5 millivolts is 60 dB.

### c. Signal Conditioners

The signal conditioners receive the analog signals from the coil preamplifiers, amplify them on the order of 30 dB, and limit signals with peak amplitudes of 7.5 volts from entering the pulse code modulation system.

### d. Pulse Code Modulation (PCM) System

The pulse code modulation system chosen for use is one designed and manufactured by Dr. Robert Lowe, Lowecom, Inc. The system features 15 channel analog input capability and offers selectable sampling rates of 2, 4, 8, 16, 32, 64, and 128 samples per second. By appropriately jumpering the analog input pins, the sampling rate may be increased by a factor of 5.

The PCM system incorporates a crystal oscillator, and associated CMOS integrated circuitry to develop the clocking pulses, and a 16 channel CMOS analog multiplexer, a 16 channel, 12 bit CMOS analog to digital converter and associated circuitry to provide the pulse coding. The crystal clock oscillator operating at a frequency of 24.576 kHz produces a square wave output with a loss rate of 1 bit in  $10^6$ . The clock pulses gate the analog multiplexer, analog to digital converter and associated follow on circuitry that form the pulse code words. The basic output is a Bi-phased pulse coded signal.

The data is organized in frames. Each frame is headed by a sync code word which is followed sequentially by the pulse coded samples from PCM channels 1 through 15. The sync code word is a pulse coded digital word with a decimal value between 0 and 4095. This word is preselected and hardwired on the circuit board. This code word is essential to the decoding process.

In the initial operation of this system, a sampling rate of 32 samples per second was utilized. Only one sample per coil per frame was analyzed, making use of PCM channels 2, 3, and 4 only.

#### e. Transmission and Recording

After the data has been PCM encoded, it is transmitted by a VHF radio link back to a receiver located at the Naval Postgraduate School, where it is currently recorded on an instrumentation tape recorder for later analysis.

### 2. Data Analysis Equipment

Currently, the recorded PCM data is played back into a PCM decoder and associated equipment which generates a nine-track 800 BPI computer tape containing the decoded sensor data. This computer tape is then input into the Naval Postgraduate School IBM 3033 computer for analysis. It is in the computer software that the sensor system transfer function is applied, spectral analysis performed, and the MAD index generated.

#### B. DATA ANALYSIS SOFTWARE

As was noted earlier, a mainframe computer was utilized to perform the spectral analysis, apply the transfer function, convert to power spectral density, plot the PSD and generate the RMS noise amplitude MAD index (using equation 4-1). This program is written in FORTRAN IV and is discussed in brief below. A copy of the program can be found in Appendix A.

The main program is divided into sections which perform the following functions:

- Data input
- Fourier analysis of time series data

- Application of system transfer function
- Projection of field components onto total field vector
- Data averaging
- Curve fitting and calculation of MAD index
- Calculation and plotting of power spectral density

### 1. Data Input

Data input is accomplished with the aid of a subroutine package supplied by Dr. Tim Stanton of the Naval Postgraduate School Department of Oceanography. His subroutine (called 'SUBROUTINE RD') serves as a FORTRAN 'READ' statement, taking the PCM data off the computer tape and converting it into integer format with a value between 0 and 4095. The data input section of the main program takes this integer value and converts it to a 'REAL' number and normalizes it to represent a voltage value between -5.0 and +5.0 volts. This section also sorts the input data matching the PCM channel to the data array representing the appropriate coil.

### 2. Fourier Analysis

The time series data is next converted to the frequency domain by utilizing a subroutine (called 'FOURT') which performs a Fast Fourier Transform (FFT). The subroutine is one available to users at the Naval Postgraduate School and utilizes the Cooley-Tukey FIT algorithm. Further information about this subroutine can be found in the program listing in Appendix A.

### 3. Application of Transfer Function and Total Field Projection

The next section of code applies the system transfer functions for the three coils to the frequency domain data.

The transfer functions are given as straight line segments which were found by least-squares approximation. The data enters this section in amplitude units of volts, and is converted into nanoTeslas by the transfer function.

Following the application of the system transfer function to the coil data, the program next calculates the projection of this data onto the earth's magnetic field vector (total field projection). This is done by first applying the local magnetic variation (declination) to the North-South (X) and East-West (Y) coil information to determine the horizontal field component. The local magnetic dip (inclination) angle is then used to project the vertical (Z-coil) and horizontal field fluctuations onto the total field fluctuation.

#### 4. Data Averaging

The previous program sections exist inside of a do-loop which enables the analysis of a long period of data without a prohibitive need for storage space. This loop includes accumulator arrays for each field component and the total field projection. The fluctuation data is converted into power prior to storage. This is done by taking the magnitude of the fluctuation component, dividing by the number of sample points, and then squaring the value. After the program passes through the averaging loop for the last time, the arithmetic average is taken for each frequency point on the arrays. At this stage the power spectrum is multiplied by the sample period to determine power spectral density.

#### 5. MAD Index Calculation

The next section computes the RMS MAD noise index previously discussed. A polynomial curve fit is performed on

the total field PSD using an available subroutine ('CHBFT'). The resulting polynomial is then integrated over the limits of the various ASQ-81 band pass settings.

#### 6. Plotting of Power Spectral Density

Plots of the power spectral density of each of the field components and the total field projection are generated in the last section of the program. This is done by converting the fluctuation power spectral density to decibels (dB) referenced to 1 nanotesla<sup>2</sup> per Hertz. A Versatec plotting subroutine ('PLOTP') is then called to actually generate the plots.

#### C. INITIAL SYSTEM OPERATION

The NPS MAD index system was initially placed into operation with the coil sensors located in the La Mesa Village housing area near the Naval Postgraduate School, Monterey, California. System checkout was accomplished in June, 1982. The full system was placed into operation on 25 July 1982 and 18 August 1982 in conjunction with similar measurements taken on the floor of Monterey Bay.

The MAD index output and power spectral density plots of the total field fluctuation for 25 July 1982, 1237-1406 local (2037-2206Z), and 18 August 1982, 0121-0250 local (0921-1050Z) and 0507-0636 local (1307-1436Z) are shown in Tables IX, X, XI, and Figures 5.4, 5.5, and 5.6 respectively.

TABLE IX

MAD Noise 2037-2206Z, 25 JUL 82, Monterey, CA

1 Gamma = 1 NanoTesla

MAD INDEX= BANDPASS:	0.1083 0.04 TO 0.20	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.1148 0.04 TO 0.40	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.1169 0.04 TO 0.60	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0888 0.06 TO 0.20	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0966 0.06 TO 0.40	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0991 0.06 TO 0.60	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0729 0.08 TO 0.20	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0822 0.08 TO 0.40	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0852 0.08 TO 0.60	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0598 0.10 TO 0.20	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0708 0.10 TO 0.40	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0743 0.10 TO 0.60	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.1234 0.04 TO 2.00	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.1122 0.06 TO 2.00	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.1018 0.08 TO 2.00	GAMMAS HERTZ
MAD INDEX= BANDPASS:	0.0923 0.10 TO 2.00	GAMMAS HERTZ

POWER SPECTRAL DENSITY OF TOTAL  
FIELD PROJECTION, LA MESA VILLAGE  
25 JUL 1982, 1237-1406 LOCAL

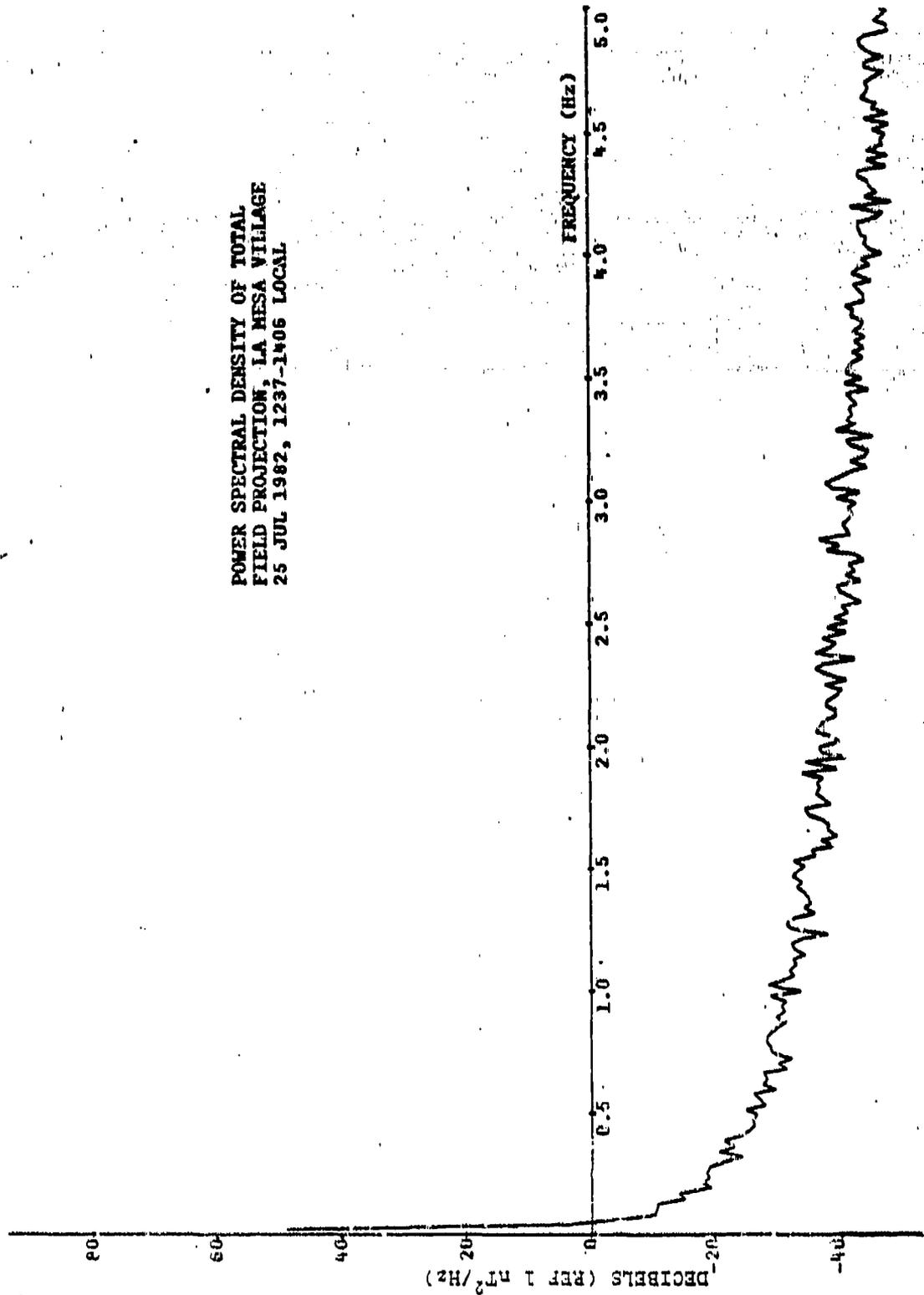


Figure 5.4 PSD 25 JUL 82, 2037-2206Z, La Mesa Village

TABLE X

MAD Noise 0921-1050Z, 18 AUG 82, Monterey, CA

1 Gamma = 1 NanoTesla

MAD INDEX= BANDPASS:	0.3644 GAMMAS 0.04 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.3960 GAMMAS 0.04 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.4004 GAMMAS 0.04 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.3027 GAMMAS 0.06 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.3401 GAMMAS 0.06 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.3453 GAMMAS 0.06 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.2528 GAMMAS 0.08 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.2966 GAMMAS 0.08 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.3025 GAMMAS 0.08 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.2118 GAMMAS 0.10 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.2624 GAMMAS 0.10 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.2691 GAMMAS 0.10 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.4047 GAMMAS 0.04 TO 2.00 HERTZ
MAD INDEX= BANDPASS:	0.3679 GAMMAS 0.06 TO 2.00 HERTZ
MAD INDEX= BANDPASS:	0.3339 GAMMAS 0.08 TO 2.00 HERTZ
MAD INDEX= BANDPASS:	0.3026 GAMMAS 0.10 TO 2.00 HERTZ

POWER SPECTRAL DENSITY OF TOTAL  
FIELD PROJECTION, LA MESA VILLAGE  
18 AUG 1982, 0121-0250 LOCAL

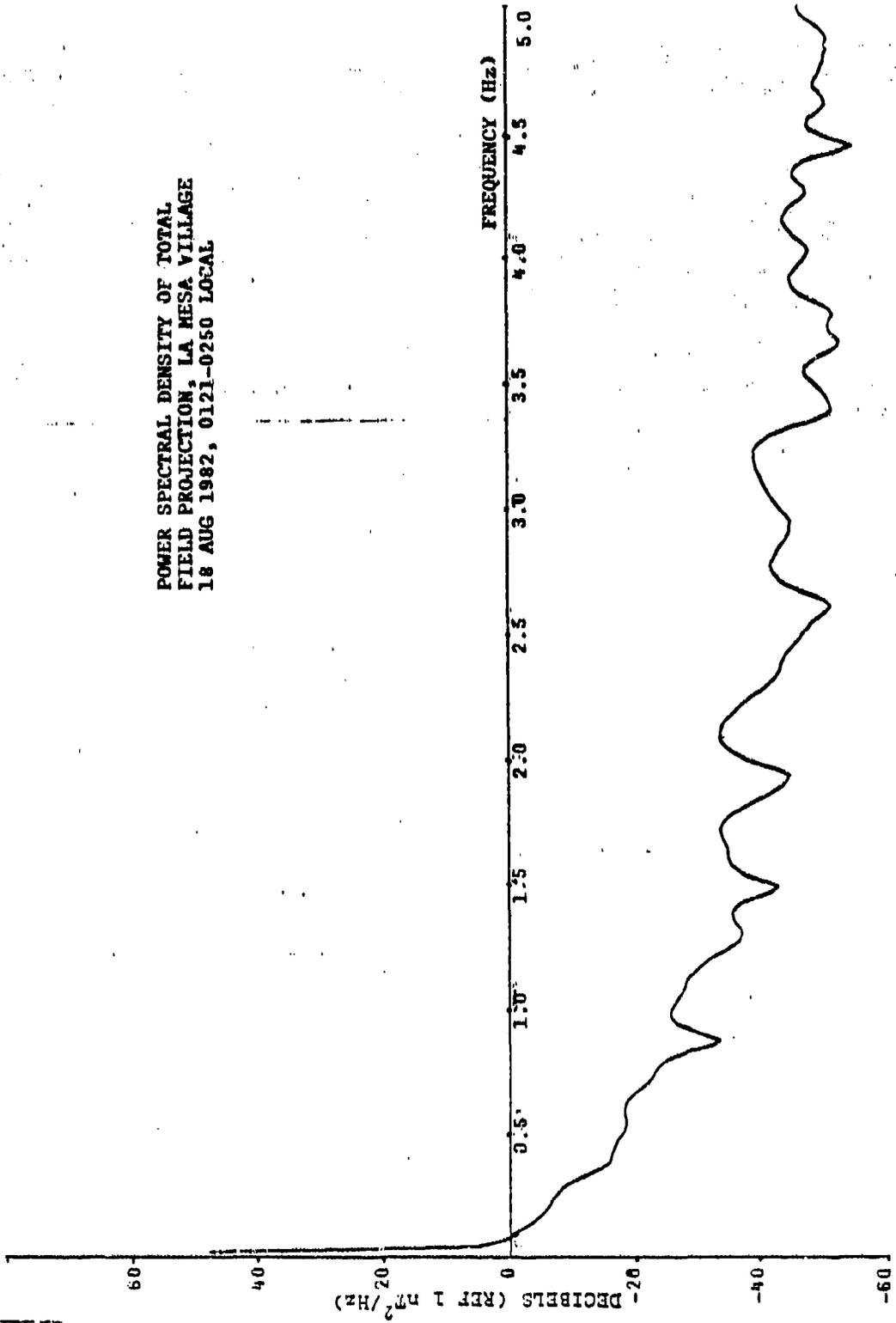


Figure 5.5 PSD 18 AUG 82, 0921-1050Z, La Mesa Village

TABLE XI

MAD Noise 1307-1436Z, 18 AUG 82, Monterey, CA

1 Gamma = 1 NanoTesla

MAD INDEX= BANDPASS:	0.3188 GAMMAS 0.04 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.3246 GAMMAS 0.04 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.3267 GAMMAS 0.04 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.2297 GAMMAS 0.06 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.2376 GAMMAS 0.06 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.2405 GAMMAS 0.06 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.1585 GAMMAS 0.08 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.1699 GAMMAS 0.08 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.1738 GAMMAS 0.08 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.1031 GAMMAS 0.10 TO 0.20 HERTZ
MAD INDEX= BANDPASS:	0.1199 GAMMAS 0.10 TO 0.40 HERTZ
MAD INDEX= BANDPASS:	0.1254 GAMMAS 0.10 TO 0.60 HERTZ
MAD INDEX= BANDPASS:	0.3504 GAMMAS 0.04 TO 2.00 HERTZ
MAD INDEX= BANDPASS:	0.3137 GAMMAS 0.06 TO 2.00 HERTZ
MAD INDEX= BANDPASS:	0.2801 GAMMAS 0.08 TO 2.00 HERTZ
MAD INDEX= BANDPASS:	0.2495 GAMMAS 0.10 TO 2.00 HERTZ

POWER SPECTRAL DENSITY OF TOTAL  
FIELD PROJECTION, LA MESA VILLAGE  
18 AUG 1982, 0507-0636 LOCAL

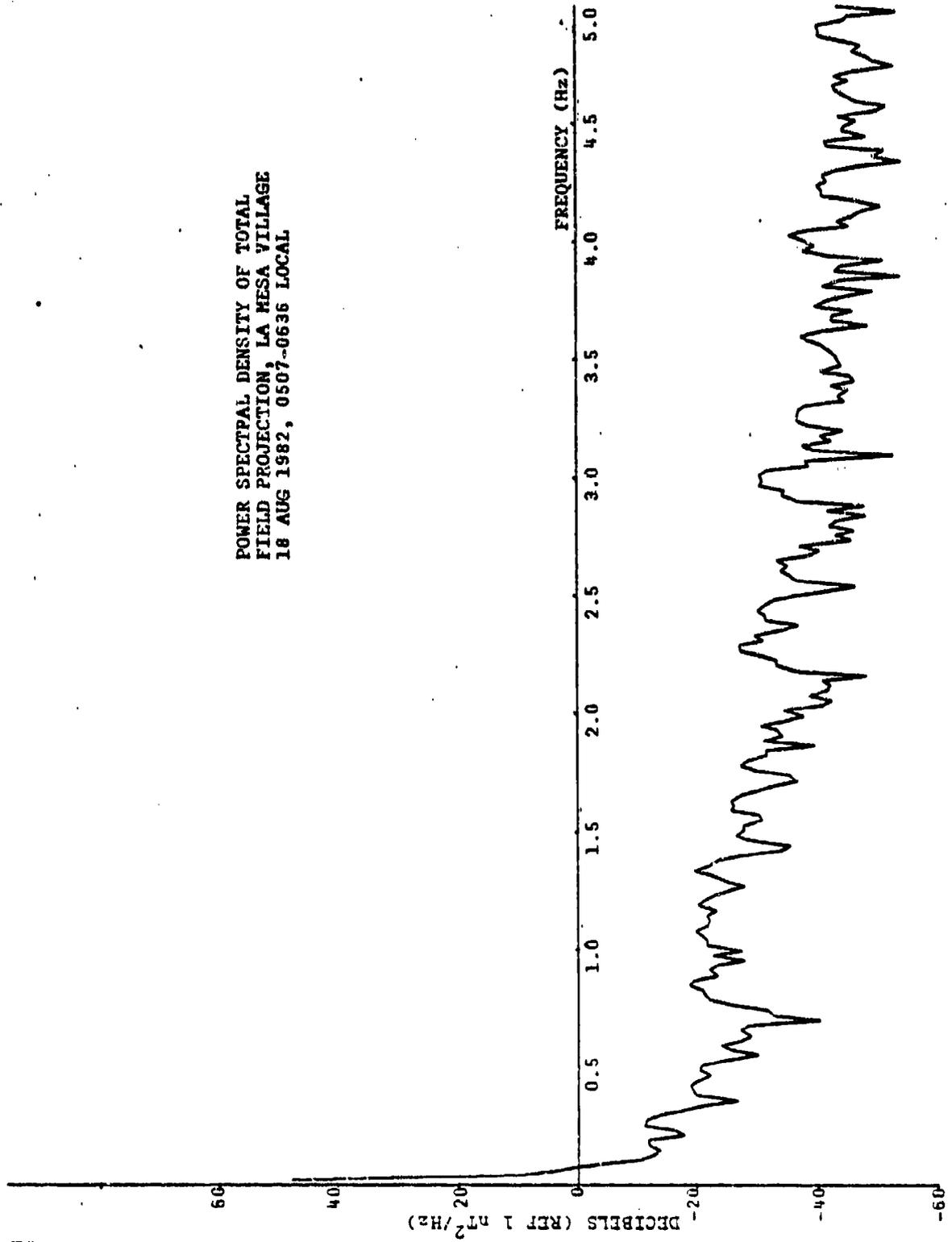


Figure 5.6 PSD 18 AUG 82, 1307-1436Z, La Mesa Village

## VI. CONCLUSION AND RECOMMENDATIONS

### A. CONCLUSION

Analysis of available information regarding current geomagnetic indices and the actual level of geomagnetic noise in the MAD bandpass indicates that the currently used indices, the K and Ap indices, are not valid for MAD operations. It is therefore desirable to derive a new index which more accurately represents the geomagnetic noise at frequencies of interest in MAD operations.

### B. RECOMMENDATIONS

A tentative index was developed, tested, and sample data obtained. The data analysis for the preliminary system was accomplished using a mainframe computer. It is recommended that further work include improving the index and the setting up of a system to on-line decode the incoming data, and utilizing a desk top computer such as the HP9845 to enable real time determination of the MAD noise index.

The spatial coherence of MAD geomagnetic noise should also be investigated, with possible application to noise cancellation. Additionally, the feasibility of the prior estimation of geomagnetic noise should be evaluated, possibly by using the prediction methods proposed by Fraser-Smith [Ref. 42, 43].

APPENDIX A

MAD INDEX DATA ANALYSIS SOFTWARE

```

MAD000060
MAD000070
MAD000080
MAD000090
MAD000100
MAD000110
MAD000120
MAD000130
MAD000140
MAD000150
MAD000160
MAD000170
MAD000180
MAD000190
MAD000200
MAD000210
MAD000220
MAD000230
MAD000240
MAD000250
MAD000260
MAD000270
MAD000280
MAD000290
MAD000300
MAD000310
MAD000320
MAD000330
MAD000340
MAD000350
MAD000360
MAD000370
MAD000380
MAD000390
MAD000400
MAD000410
MAD000420
MAD000430
MAD000440
MAD000460
MAD000470
MAD000480
MAD000490
MAD000500
MAD000510
MAD000520
MAD000530

//JEFF$1L JOB (3224,0129),SCHWEIGER SMC 2239,CLASS=F
//*MAIN ORG=NPGVMI,3224P,LINES=(10)
//*FCRMAT PR,DDNAME=PLOT,SYSVECTR,DEST=LOCAL
// EXEC FRTXCLGP,PARM.LKED=LIST,MAP,XREF,REGION.GO=1200K
//FCRT. SYSIN DD *
INTEGER*2 IN(16)
ARRAY IN IS USED IN READING DATA FROM TAPE
COMPLEX*8 XX(9500),YY(9500),ZZ(9500),TF(9500)
THE COMPLEX*8 ARRAYS ARE USED TO ORDER INPUT DATA AND INITIALLY
REPRESENT VOLTAGE - TIME SERIES INFORMATION.
DIMENSION TIME(9500),FREQ(9500),WOK(19000),FRQ2(500)
DIMENSION ZX(9500),ZY(9500),ZV(9500),ZT(9500)
THE *Z* ARRAYS REPRESENT FREQUENCY DOMAIN (FF TRANSFORMED)
MAGNITUDE DATA AND ARE EVENTUALLY CONVERTED TO POWER SPECTRAL
DENSITY INFORMATION.
INTEGER R(10)
DIMENSION RX(10),PX(94),PY(94),PC(10),PD(9),PX(40),PY(40)
THE *R* AND *P* ARRAYS ARE USED IN CURVE FITTING THE
POWER SPECTRAL DENSITY OF THE ACTIVITY PROJECTION ON THE
TOTAL FIELD.
DIMENSION FL(4),FU(4)
INTEGER*4 ITB(12)/12*0/
REAL*4 RTB(28)/28*0.0/
REAL ALAB(4)/CH-X,CH-Y,CH-Z, TOT//
REAL*8 TITLE(12)
EQUIVALENCE(TITLE(1),RTB(5))
ARRAYS TITLE,RTB,ALAB,AND *TITLE* ARE USED IN GENERATING
THE VERTSA TEC PLOTTER OUTPUT.
DATA XX,YY,ZZ,TF/38000*(0.,0.)/
DATA ZX,ZY,ZV,ZT/38000*0.0/
DATA FL,FC4,06,08,1/
DATA FU,02,04,06,2./
DATA TIME,FREQ/19000*0.0/
DATA PX,PY,PXI,PYI/268*0.0/
TWOPI=6.2831853
COS60=COS(TWOPI/6.)
COS30=COS(TWOPI/12.)
D=16.75*TWOPI/360.
COSD=COS(D)
COSDI=COS(90.-D)*TWOPI/360.
D IS THE DECLINATION OR MAGNETIC VARIATION AT THE MAGNETOMETER
SITE.
THE NEXT FIVE LINES SERVE AS A TIME DELAY IN STARTING THE
DATA ANALYSIS
ISEC=3
ITL=ISEC*32
DC 55 JJ=1,ITL
CALL RD(20,IN,1000,IREC,IRR)

```

MAD00540  
MAD00550  
MAD00560  
MAD00570  
MAD00580  
MAD00590  
MAD00600  
MAD00610  
MAD00620  
MAD00630  
MAD00640  
MAD00650  
MAD00660  
MAD00670  
MAD00680  
MAD00690  
MAD00700  
MAD00710  
MAD00720  
MAD00730  
MAD00740  
MAD00750  
MAD00760  
MAD00770  
MAD00780  
MAD00790  
MAD00800  
MAD00810  
MAD00820  
MAD00830  
MAD00840  
MAD00850  
MAD00860  
MAD00870  
MAD00880  
MAD00890  
MAD00900  
MAD00910  
MAD00920  
MAD00930  
MAD00940  
MAD00950  
MAD00960  
MAD00970  
MAD00980  
MAD00990  
MAD01000  
MAD01010

```

55 CONTINUE
   IFRAME=9500
   NR=18
   FN=FLOAT(NR)
   DC 70 L1=1, NR
   THE DO LOOP ENDING WITH STATEMENT 70 ENABLES THE PROGRAM TO
   PROCESS A LARGE AMOUNT OF DATA BY REPEATING THE PROCESS IN
   BLOCKS. THE DATA POINTS FROM EACH RUN THROUGH THE DC LOOP ARE
   ADDED TOGETHER AND EVENTUALLY AVERAGED BY THE NUMBER OF RUNS
   THROUGH THE DO LOOP.
   'NR' REPRESENTS THE NUMBER OF DATA SEQUENCES TO BE AVERAGED.
   I SEQUENCE CURRENTLY EQUALS 9500 DATA POINTS FOR EACH CHANNEL
   OR 296.88 SECONDS OF DATA.
   THE DO LOOP ENDING WITH 60 READS THE DATA FROM THE PCM FRAME
   STRIPS OUT THE SYNC CODE, AND SORTS OUT THE DATA BY COIL
   CHANNEL
   DC 60 JJ=1, IFRAME
   CALL RD(20, IN, 1000, IREC, IRR)
   XX(JJ)=IN(2)
   YY(JJ)=IN(3)
   ZZ(JJ)=IN(4)
60 CONTINUE
   WRITE(6, 200) IRR, IREC
   FORMAT(10X, 'IRR=', I6, ' IREC=', I6, '/')
   THE FOLLOWING SECTION GENERATES THE TIME AND FREQUENCY
   ARRAYS AND NORMALIZES THE INPUT PCM DATA TO VOLTAGE FORM
   IN PREPARATION FOR FAST FOURIER TRANSFORM TO THE FREQUENCY
   DOMAIN.
   N=9500
   FN=FLCAT(N)
   DELTAT=1./32.
   T=FN*DELTAT
   DELTAF=TWCP I/(FN*DELTAT)
   DC 20 J=1, DELTAT*FLCAT(J)
   FREQ(J)=DELTAF*FLCAT(J)
   XX(J)=(XX(J)-2048.)*5./2048.
   YY(J)=REAL(XX(J))
   YY(J)=REAL(YY(J))-2048.)*5./2048.
   ZZ(J)=REAL(ZZ(J))-2048.)*5./2048.
   'XX' IS THE X-COIL DATA, 'YY' IS THE Y-COIL DATA,
   'ZZ' IS THE Z-COIL DATA, AND 'TF' IS THE PROJECTION OF THE
   NCRTH-SOUTH COMPONENT (XX) AND THE VERTICAL COMPONENT (ZZ)
   ON THE TOTAL GEOMAGNETIC FIELD VECTOR.
20 CONTINUE

```

MAD01020  
MAD01030  
MAD01040  
MAD01050  
MAD01060  
MAD01070  
MAD01080  
MAD01090  
MAD01100  
MAD01110  
MAD01120  
MAD01130  
MAD01140  
MAD01150  
MAD01160  
MAD01170  
MAD01180  
MAD01190  
MAD01200  
MAD01210  
MAD01220  
MAD01230  
MAD01240  
MAD01250  
MAD01260  
MAD01270  
MAD01280  
MAD01290  
MAD01300  
MAD01310  
MAD01320  
MAD01330  
MAD01340  
MAD01350  
MAD01360  
MAD01370  
MAD01380  
MAD01390  
MAD01400  
MAD01410  
MAD01420  
MAD01430  
MAD01440  
MAD01450  
MAD01460  
MAD01470  
MAD01480  
MAD01490

```

DO 21 J=1,500
FRQ2(J)=ALOG10(FREQ(J))
21 CONTINUE
C TIME SERIES DATA. PERFORM AN FFT ON THE INPUT
C FURTHER INFORMATION. SEE THE WRITEUP ON 'FOUR' FOR
C CALL FOURT(XX,N,1,-1,0,WORK)
C CALL FOURT(YY,N,1,-1,0,WORK)
C THE NEXT BLOCK OF STATEMENTS APPLY THE SYSTEM (VOLTAGE TO
B-FIELD) TRANSFER FUNCTION TO THE TRANSFORM S.
C GAMMA IN DATA. THIS BLOCK ENDS AT STATEMENT 5.
C THE TRANSFER FUNCTION CONVERTS VOLTS TO NANOTESLAS (GAMMAS).
C **WARNING** THIS TRANSFER FUNCTIONS YIELDS AN INACCURATE
C PHASE. USE A DIFFERENT TRANSFER FUNCTION IF PHASE INFORMATION
C NEEDED.
DO 9 L=1,N
FPQ=FREQ(L)
1 IF(FRQ.LE.25.160 TO 1
XX(L)=XX(L)/28.
GO TO 8
1 IF(FRQ.LE.15.160 TO 2
XX(L)=XX(L)/(105.5-3.14*FRQ)
YY(L)=YY(L)/(191.32-7.588*FRQ)
ZZ(L)=ZZ(L)/(177.26-7.484*FRQ)
GO TO 8
2 IF(FRQ.LE.10.160 TO 3
XX(L)=XX(L)/(5.958*FRQ-30.97)
YY(L)=YY(L)/(7.166*FRQ-39.99)
ZZ(L)=ZZ(L)/(6.49*FRQ-32.35)
GO TO 8
3 IF(FRQ.LE.7.5160 TO 4
XX(L)=XX(L)/(3.492*FRQ-6.31)
YY(L)=YY(L)/(4.252*FRQ-10.85)
ZZ(L)=ZZ(L)/(4.044*FRQ-7.89)
GO TO 8
4 IF(FRQ.LE.5.160 TO 5
XX(L)=XX(L)/(2.6311*FRQ+0.14667)
YY(L)=YY(L)/(3.012*FRQ-1.55)
ZZ(L)=ZZ(L)/(3.184*FRQ-1.44)
GO TO 8
5 IF(FRQ.LE.3.160 TO 6
XX(L)=XX(L)/(2.6311*FRQ+0.14667)
YY(L)=YY(L)/(2.702*FRQ)
ZZ(L)=ZZ(L)/(2.92*FRQ)
GO TO 8
6 XX(L)=XX(L)/(2.72*FRQ)
GO TO 7

```

```

MAD01500
MAD01510
MAD01520
MAD01530
MAD01540
MAD01550
MAD01560
MAD01570
MAD01580
MAD01590
MAD01600
MAD01610
MAD01620
MAD01630
MAD01640
MAD01650
MAD01660
MAD01670
MAD01680
MAD01690
MAD01700
MAD01710
MAD01720
MAD01730
MAD01740
MAD01750
MAD01760
MAD01770
MAD01780
MAD01790
MAD01800
MAD01810
MAD01820
MAD01830
MAD01840
MAD01850
MAD01860
MAD01870
MAD01880
MAD01890
MAD01900
MAD01910
MAD01920
MAD01930
MAD01940
MAD01950
MAD01960
MAD01970

6 CONTINUE
  TP(L)=(XX(L)*COSD + Y(L)*COSD1)*COS60 + ZZ(L)*COS30
5 CONTINUE
  IN THE NEXT BLOCK, ENDING WITH 30, THE MAGNITUDE OF THE COMPLEX
  FREQUENCY DOMAIN DATA IS TAKEN AND PLACED IN A REAL ARRAY. *ZX*
  FOR THE X-COIL, *ZY* FOR THE Y-COIL, *ZV* FOR THE Z OR VERTICAL
  COIL AND *ZT* FOR THE TOTAL FIELD PROJECTION. THIS DATA IS
  DIVIDED BY THE NUMBER OF SAMPLE POINTS AND SQUARED TO DETERMINE
  POWER.
  DO 30 I1=1,N
    ZX(I1)=ZX(I1)/(FN)**2
    ZY(I1)=ZY(I1)/(FN)**2
    ZV(I1)=ZV(I1)/(FN)**2
    ZT(I1)=ZT(I1)/(FN)**2
  CCNTINUE
30 CONTINUE
  THE FOLLOWING BLOCK AVERAGES THE DATA POINTS ADDED IN BLOCK 30
  ABOVE BY THE NUMBER OF RUNS THROUGH THE DO LOOP ENDING WITH 70.
  AT THIS POINT THE POWER CURVES ARE CONVERTED INTO POWER SPECTRAL
  DENSITY.
  DO 33 I3=1,M3
    ZX(I3)=ZX(I3)*T/FNR
    ZY(I3)=ZY(I3)*T/FNR
    ZV(I3)=ZV(I3)*T/FNR
    ZT(I3)=ZT(I3)*T/FNR
  CCNTINUE
  THE NEXT STATEMENTS, THROUGH 204, ARE USED IN CURVE FITTING
  THE POWER SPECTRAL DENSITY OF THE TOTAL FIELD PROJECTION
  AND IN CALCULATING A MAD ACTIVITY INDEX.
  DO 37 I=3,42
    PX(I-2)=FREC(I)
    PY(I-2)=ZT(I)
  CCNTINUE
  M2=8
  M3=9
  WRITE(6,201)
  FORMAT(I1)
  CALL CHBFT(PX,PY,40,PC,M2,RX,RH,R)
  DO 38 I=1,M3
    PD(I)=PC(I)/FLOAT(I)
  CCNTINUE
  DO 39 K=1,4
    DO 46 LL=1,2
      XL=0.
  DC 47 I=1,9
  XU=XU+PD(I)*{FU(L)**I}
  XL=XL+PD(I)*{FL(K)**I}

```

MAD01980  
MAD01990  
MAD02000  
MAD02010  
MAD02020  
MAD02030  
MAD02040  
MAD02050  
MAD02060  
MAD02070  
MAD02080  
MAD02090  
MAD02100  
MAD02110  
MAD02120  
MAD02130  
MAD02140  
MAD02150  
MAD02160  
MAD02170  
MAD02180  
MAD02190  
MAD02200  
MAD02210  
MAD02220  
MAD02230  
MAD02240  
MAD02250  
MAD02260  
MAD02270  
MAD02280  
MAD02290  
MAD02300  
MAD02310  
MAD02320  
MAD02330  
MAD02340  
MAD02350  
MAD02360  
MAD02370  
MAD02380  
MAD02390  
MAD02400  
MAD02410  
MAD02420  
MAD02430  
MAD02440  
MAD02450

```

47 CONTINUE
IF(XU.LE.XL 160 TO 71
GO TO 72
71 M2=M2-1
M3=M3-1
CALL CHBFT(PX,PY,40,PC,M2,RX,RH,R)
DO 73 I=1,M2
PD(I)=PC(I)/FLOAT(I)
73 CONTINUE
XU=0.
XL=0.
DO 74 I=1,M2
XU=XU+PD(I)* (FU(LL)**I)
XL=XL+PD(I)* (FL(K)**I)
74 CONTINUE
72 XHI=SQRT(XU-XL)
WRITE(6,204)XHI,FL(K),FU(LL)
204 FORMAT(10X,'MAG INDEX= ',F10.4,1X,'GAMMAS',/,10X,'B ANDPASS:',2X,
F5.2,/, TO ,F5.2,' HERTZ',/)
45 CONTINUE
39 DO 31 I=3,96
PX(I-2)=FREQ(I)
PY(I-2)=ZT(I)
31 CONTINUE
M2=8
M3=9
CALL CHBFT(PX1,PY1,94,PC,M2,RX,RH,R)
DO 34 I=1,M3
PD(I)=PC(I)/FLOAT(I)
34 CONTINUE
DO 35 K=1,4
XU=0.
XL=0.
DO 36 I=1,9
XU=XU+PD(I)* (FU(4)**I)
XL=XL+PD(I)* (FL(K)**I)
36 CONTINUE
IF(XU.LE.XL 160 TO 75
GO TO 76
75 M2=M2-1
M3=M3-1
CALL CHBFT(PX1,PY1,94,PC,M2,RX,RH,R)
DO 77 I=1,M3
PD(I)=PC(I)/FLOAT(I)
IM=I-1
77 CONTINUE
XU=0.

```

MAD02460  
MAD02470  
MAD02480  
MAD02490  
MAD02500  
MAD02510  
MAD02520  
MAD02530  
MAD02540  
MAD02550  
MAD02560  
MAD02570  
MAD02580  
MAD02590  
MAD02600  
MAD02610  
MAD02620  
MAD02630  
MAD02640  
MAD02650  
MAD02660  
MAD02670  
MAD02680  
MAD02690  
MAD02700  
MAD02710  
MAD02720  
MAD02730  
MAD02740  
MAD02750  
MAD02760  
MAD02770  
MAD02780  
MAD02790  
MAD02800  
MAD02810  
MAD02820  
MAD02830  
MAD02840  
MAD02850  
MAD02860  
MAD02870  
MAD02880  
MAD02890  
MAD02900  
MAD02910  
MAD02920  
MAD02930

```

XL=0.      I=1, M3
DC 78      XU=XL+PD(I)*(FU(4)**I)
           XL=XL+PD(I)*(FL(K)**I)
78 CONTINUE
76 XMI=SQRT(XU-XL)
   WRITE(6,204)XMI,FL(K),FU(4)
35 CONTINUE
   THE NEXT DO LOOP CONVERTS THE B-FIELD POWER SPECTRAL DENSITY
   INTO DECIBELS REFERENCED TO 1 NANOTESLA (GAMMA) **2/HERTZ.
   32 I=1,N
   ZX(I)=10.*ALOG10(ZX(I))
   ZY(I)=10.*ALOG10(ZY(I))
   ZV(I)=10.*ALOG10(ZV(I))
   ZT(I)=10.*ALOG10(ZT(I))
32 CONTINUE

C VERSATEC PLOT OF B - FIELD SPECTRA
C
NPTS=10./DELTA F +1.
NPD2=NPTS/2 +1
* NPTS, DETERMINES NUMBER OF POINTS NECESSARY IN ORDER FOR
  THE 0 TO 10 HERTZ RANGE TO BE PLOTTED.
* NPD2, IS FOR THE 0 TO 5 HERTZ RANGE.
  FOR THE FOLLOWING 'ITB' AND 'RTB' VALUES REVIEW THE WRITE-UP
  FOR THE SUBROUTINE PROCEDURE 'CRAMP'.
ITB(3)=20
ITB(4)=8
ITB(12)=0
RTB(1)=0.0
RTB(2)=0.0
RTB(3)=ALAB(1)
READ(5,300)ITILE
CALL DRAWP(NPTS,FRQ2,ZX,ITB,RTB)
RTB(3)=ALAB(2)
READ(5,300)ITILE
CALL DRAWP(NPTS,FRQ2,ZY,ITB,RTB)
RTB(3)=ALAB(3)
READ(5,300)ITILE
CALL DRAWP(NPTS,FRQ2,ZV,ITB,RTB)
ITB(3)=10
ITB(12)=1
RTB(3)=ALAB(4)
READ(5,300)ITILE
CALL DRAWP(NPD2,FRQ2,ZT,ITB,RTB)
3000 FORMAT(6A8)
      STOP
      END

```



MAD03420  
MAD03430  
MAD03440  
MAD03450  
MAD03460  
MAD03470  
MAD03480  
MAD03490  
MAD03500  
MAD03510  
MAD03520  
MAD03530  
MAD03540  
MAD03550  
MAD03560  
MAD03570  
MAD03580  
MAD03590  
MAD03600  
MAD03610  
MAD03620  
MAD03630  
MAD03640  
MAD03650  
MAD03660  
MAD03670  
MAD03680  
MAD03690  
MAD03700  
MAD03710  
MAD03720  
MAD03730  
MAD03740  
MAD03750  
MAD03760  
MAD03770  
MAD03780  
MAD03790  
MAD03800  
MAD03810  
MAD03820  
MAD03830  
MAD03840  
MAD03850  
MAD03860  
MAD03870  
MAD03880  
MAD03890

```

100 CONTINUE
C IF (IRR.EQ.0) GO TO 150
IRR=IRR+1
IF (IRR.LT.IRS) GO TO 120
WRITE (6,110)
FORMAT ('I STOPPED IN SUB RD BECAUSE OF IRR.GT.',I6,' AT L110')
110 IRR=IRR
STOP
CONTINUE
WRITE (6,130) IREC,IRR
FORMAT (' RESYNC AT FRAME ',I6,' WITH TOTAL ERRORS ',I7)
130 IERR=0
IRG=IRR
GO TO 50
CONTINUE
RETURN
150 WRITE (6,910) IUN,IREC
FORMAT ('I END CF UNIT ',I3,' AT REC ',I7)
STOP
END

FUNCTION ISHIFT (IN,NPLC)
RETURN S HIFTED VALUE OF I*2 WORD IN
--VE LEFT,+VE RIGHT SHIFT

INTEGER * 2 IN
IP=IN
IF (IP.LT.0) IP=IP+65536
IF (NPLC.LT.0) GC TO 30
ISHIFT=IP/(2**IABS(NPLC))
RETURN
ISHIFT=IP*(2**IABS(NPLC))
IF (ISHIFT.GT.65535) ISHIFT=MOD(ISHIFT,65536)
RETURN
END
FUNCTION IMASK (IN,IBL,IBR)
MASK I*2 WORD IN OUTSIDE BITS IBL & IBR

INTEGER * 2 IN,IO
IO=IN
IF (IBR.EQ.0) GO TO 50
IT=ISHIFT(IN,IBR)
IO=IT
IP=ISHIFT(IO,IBL-15-IBR)
IO=IP
IMASK=ISHIFT(IO,15-IBL)
RETURN
50

```

MAD03900  
MAD03910  
MAD03920  
MAD03930  
MAD03940  
MAD03950  
MAD03960  
MAD03970  
MAD03980  
MAD03990  
MAD04000  
MAD04010  
MAD04020  
MAD04030  
MAD04040  
MAD04050  
MAD04060  
MAD04070  
MAD04080  
MAD04090  
MAD04100  
MAD04110  
MAD04120  
MAD04130  
MAD04140  
MAD04150  
MAD04160  
MAD04170  
MAD04180  
MAD04190  
MAD04200  
MAD04210  
MAD04220  
MAD04230  
MAD04240  
MAD04250  
MAD04260  
MAD04270  
MAD04280  
MAD04290  
MAD04300  
MAD04310  
MAD04320  
MAD04330  
MAD04340  
MAD04350  
MAD04360  
MAD04370

```

END
SUBROUTINE FOURT
PURPOSE
SUBROUTINE FOURT COMPUTES THE FORWARD AND INVERSE
COOLEY-TUKEY FAST FOURIER TRANSFORM OF THE CONTENTS OF THE
ARRAY DATA. FOR DATA A SINGLY-DIMENSIONED ARRAY OF LENGTH
L, THE JTH COMPONENT OF THE TRANSFORM IS GIVEN BY
SUM(DATA(K)*W**((K-1)*(J-1)))
WHERE THE SUM IS TAKEN OVER K=1..LE. K=LE, L, AND
W=EXP(ISIGN*2*PI*SORT(-1)/L)

THE VALUE OF ISIGN DEPENDS UPON WHETHER A FORWARD OR INVERSE
TRANSFORM IS TO BE PERFORMED. FOURT MAY ALSO BE USED ON A
MULTI-DIMENSIONAL ARRAY, IN WHICH CASE A FOURIER TRANSFORM IS
PERFORMED ALONG EACH DIMENSION IN TURN.

CALLING SEQUENCE
CALL FOURT(DATA,NN,NDIM,ISIGN,IFORM,WORK)

DESCRIPTIGN OF ARGUMENTS
DATA COMPLEX*8 MULTI-DIMENSIONAL ARRAY CONTAINING THE DATA TO
BE TRANSFORMED. ON OUTPUT DATA CONTAINS THE TRANSFORM.
NORMAL FORTRAN ORDERING IS EXPECTED, THE FIRST SUBSCRIPT
CHANGING THE FASTEST.

AN INTEGER*4 ARRAY CONTAINING THE DIMENSIONS OF THE ARRAY
DATA.

NDIM NUMBER OF DIMENSIONS OF THE ARRAY DATA = NUMBER OF ELE-
MENTS IN THE ARRAY NN.

ISIGN INTEGER INDICATING WHETHER FORWARD OR INVERSE TRANSFORM
IS TO BE PERFORMED.
ISIGN=-1 FOR FORWARD TRANSFORM
ISIGN=1 FOR INVERSE TRANSFORM.
NOTE: THESE DEFINITIONS ARE NOT STANDARDIZED. IN PAR-
TICULAR, THE DEFINITIONS OF FORWARD AND INVERSE
TRANSFORM ARE REVERSED IN THE IMSL FFT ROUTINES.

IFORM AN INTEGER INDICATING WHETHER OR NOT DATA CONTAINS ONLY
PURELY REAL VALUES.
IFORM=0 IF DATA IS PURELY REAL
IFORM=1 OTHERWISE.
IF IFORM IS SET TO 0, ALL THE IMAGINARY PARTS OF THE ELE-

```

MAD04380  
MAD04390  
MAD04400  
MAD04410  
MAD04420  
MAD04430  
MAD04440  
MAD04450  
MAD04460  
MAD04470  
MAD04480  
MAD04490  
MAD04500  
MAD04510  
MAD04520  
MAD04530  
MAD04540  
MAD04550  
MAD04560  
MAD04570  
MAD04580  
MAD04590  
MAD04600  
MAD04610  
MAD04620  
MAD04630  
MAD04640  
MAD04650  
MAD04660  
MAD04670  
MAD04680  
MAD04690  
MAD04700  
MAD04710  
MAD04720  
MAD04730  
MAD04740  
MAD04750  
MAD04760  
MAD04770  
MAD04780  
MAD04790  
MAD04800  
MAD04810  
MAD04820  
MAD04830  
MAD04840  
MAD04850

MENTS IN DATA MUST BE SET TO 0.0.

WORK A 1-DIMENSIONAL REAL\*4 ARRAY USED FOR WORKING STORAGE.  
ITS LENGTH SHOULD BE THREE TIMES THE LARGEST ARRAY DIMENSION  
NN(I), I=1,2,3,4,5,6,7,8,9,10,11,12,13,14,15,16,17,18,19,20,21,22,23,24,25,26,27,28,29,30,31,32,33,34,35,36,37,38,39,40,41,42,43,44,45,46,47,48,49,50,51,52,53,54,55,56,57,58,59,60,61,62,63,64,65,66,67,68,69,70,71,72,73,74,75,76,77,78,79,80,81,82,83,84,85,86,87,88,89,90,91,92,93,94,95,96,97,98,99,100.  
PARTICULAR, IF ALL NN(I) ARE POWERS OF TWO, NO WORK SPACE  
IS NEEDED AND WORK MAY BE REPLACED BY ZERO IN THE CALLING  
SEQUENCE.

REMARKS

IF AN INVERSE TRANSFORM (ISIGN=+1) IS PERFORMED UPON AN ARRAY  
OF TRANSFORMED (ISIGN=-1) DATA, THE ORIGINAL DATA WILL REAP-  
PEAR, MULTIPLIED BY NN(I)\*NN(2)\*...\*NN(NDIM).

FOR A MULTI-DIMENSIONAL ARRAY THE (J1,J2,...,JNDIM)  
COMPONENT OF THE TRANSFORM IS GIVEN BY  
SUM(DATA(I1,I2,...,INDIM4)\*W1\*\*((I1-1)\*(J1-1))\*  
W2\*\*((I2-1)\*(J2-1))\*...\*WINDIM\*\*((INDIM-1)\*(JNDIM-1))  
HERE THE SUM RANGES OVER ALL POSSIBLE VALUES OF THE I'S  
AND W1=EXP(ISIGN\*2\*PI\*SQRT(-1)/NN(I)), ETC.

THE ARRAY OF INPUT DATA MUST BE IN COMPLEX FORMAT. THE DATA  
HOWEVER, IS ALL IMAGINARY PARTS ARE ZERO (I.E. FORTY PER-  
CENT). (FOR FASTEST TRANSFORM OF REAL DATA, NN(I) SHOULD BE E-  
VEN.) THE TRANSFORM VALUES ARE ALWAYS COMPLEX AND ARE RETURNED  
IN THE ORIGINAL ARRAY OF DATA, REPLACING THE INPUT DATA. INTEGERS.  
LENGTH OF EACH DIMENSION OF THE DATA ARRAY MAY BE ANY INTEGER,  
AND IS PARTICULARLY FAST ON COMPOSITE INTEGERS THAN ON PRIMES,  
AND IS PARTICULARLY FAST ON NUMBERS RICH IN FACTORS OF TWO.

TIMING IS IN FACT GIVEN BY THE FOLLOWING FORMULA. LET NTOI BE  
THE TOTAL NUMBER OF POINTS (REAL OR COMPLEX) IN THE DATA ARRAY,  
THAT IS, NTOI=NN(1)\*NN(2)\*...\*NN(NDIM). LET SUM2 BE THE  
SUM OF THE FACTORS OF TWO IN NTOI THAT IS, SUM2 = 2\*\*K2. LET  
SUMF BE THE SUM OF ALL OTHER FACTORS OF NTOI THAT IS, SUMF =  
3\*\*K3\*5\*\*K5\*... DATA IS T = T0 + NTOI\*(T1+T2\*SUM2+T3\*SUMF). T = 3000+  
CDC 3300 (FLOATING POINT ADD TIME = SIX MICROSECONDS). T = 3000+  
NTOI\*(600+40\*SUM2+175\*SUMF) MICROSECONDS ON COMPLEX DATA.

THE SAVINGS OFFERED BY THIS PROGRAM CAN BE DRAMATIC. A ONE-DI-  
MENSIONAL ARRAY 4000 IN LENGTH WILL BE TRANSFORMED IN 4000\*(600+  
40\*(2+2+2+2+2)+175\*(5+5+5)) = 14.5 SECONDS VERSUS ABOUT 4000\*  
4000\*175 = 2800 SECONDS FOR THE STRAIGHTFORWARD TECHNIQUE.

CC



MAD05340  
MAD05350  
MAD05360  
MAD05370  
MAD05380  
MAD05390  
MAD05400  
MAD05410  
MAD05420  
MAD05430  
MAD05440  
MAD05450  
MAD05460  
MAD05470  
MAD05480  
MAD05490  
MAD05500  
MAD05510  
MAD05520  
MAD05530  
MAD05540  
MAD05550  
MAD05560  
MAD05570  
MAD05580  
MAD05590  
MAD05600  
MAD05610  
MAD05620  
MAD05630  
MAD05640  
MAD05650  
MAD05660  
MAD05670  
MAD05680  
MAD05690  
MAD05700  
MAD05710  
MAD05720  
MAD05730  
MAD05740  
MAD05750  
MAD05760  
MAD05770  
MAD05780  
MAD05790  
MAD05800  
MAD05810

```

CALL FOURT(DATA,NN,3,-1,1,WORK)
EXAMPLE 2. ONE-DIMENSIONAL FORWARD TRANSFORM OF A REAL ARRAY OF
LENGTH 64 IN FORTRAN II
DIMENSION DATA(2,64)
DO 2 I=1,64
DATA(I,1)=REAL PART
DATA(2,I)=0.
CALL FOURT(DATA,64,1,-1,0,0)
PROGRAMMER
PROGRAM BY NORMAN BRENNER FROM THE BASIC PROGRAM BY CHARLES
RADER, JUNE 1967. THE IDEA FOR THE DIGIT REVERSAL WAS
SUGGESTED BY RALPH ALTER.
DOCUMENTATION REVISED BY JOANNE BCGART, AUGUST 1979, NPS.
SUBROUTINE FOURT(DATA,NN,NDIM,ISIGN,IFORM,WORK)
DIMENSION DATA(1),NN(1),IFACT(32),WORK(1)
DATA TWOP1/6.28318530717967,P.THCF/0.70710678118655/
IF(NDIM-1)920,1,1
NTOT=2
DC 2 IDIM=1,NDIM
IF(NN(IDIM))920,920,2
NTOT=NTOT*NN(IDIM)
MAIN LGOP FOR EACH DIMENSION
NP1=2
DO 910 IDIM=1,NDIM
N=NA(IDIM)
NP2=NP1*N
IF(N-1)920,500,5
IS N A POWER OF TWO AND IF NOT, WHAT ARE ITS FACTORS
M=N
NTWC=NP1
IF=1
IDIV=2
IQOUT=M/IDIV
IREM=M-IDIV*IQOUT
IF(IQOUT-IDIV)50,11,11
IF(IREM)20,12,20
NTWC=NTWC+NTWO
IFACT(IF)=IDIV
IF=IF+1
M=IQOUT

```

2  
1  
2  
C  
C  
C  
5  
10  
11  
12

MAD05820  
MAD05830  
MAD05840  
MAD05850  
MAD05860  
MAD05870  
MAD05880  
MAD05890  
MAD05900  
MAD05910  
MAD05920  
MAD05930  
MAD05940  
MAD05950  
MAD05960  
MAD05970  
MAD05980  
MAD05990  
MAD06000  
MAD06010  
MAD06020  
MAD06030  
MAD06040  
MAD06050  
MAD06060  
MAD06070  
MAD06080  
MAD06090  
MAD06100  
MAD06110  
MAD06120  
MAD06130  
MAD06140  
MAD06150  
MAD06160  
MAD06170  
MAD06180  
MAD06190  
MAD06200  
MAD06210  
MAD06220  
MAD06230  
MAD06240  
MAD06250  
MAD06260  
MAD06270  
MAD06280  
MAD06290

```

20 GO TO 10
30 IDIV=3
   IF INON2=IF
   IQUOT=M/IDIV*IGUOT
   IREM=M-IDIV*IGUOT
   IF(IQUOT-IDIV)60,31,31
   IF(IREM)40,22,40
   IFACT(IF)=ICIV
   IF=IF+1
   M=IGUOT
   GO TO 30
40 IDIV=IDIV+2
   GO TO 30
50 INON2=IF
   IF(IREM)60,51,60
   NTWO=NTWO+NTWO
   GO TO 70
60 IFACT(IF)=M

SEPARATE FOUR CASES---
1. COMPLEX TRANSFORM OR REAL TRANSFORM FOR THE 4TH, 9TH, ETC.
   DIMENSIONS.
2. REAL TRANSFORM FOR THE 2ND OR 3RD DIMENSION. METHCD--
   TRANSFORM HALF THE DATA, SUPPLYING THE OTHER HALF BY CON-
   JUGATE SYMMETRY.
3. REAL TRANSFORM FOR THE 1ST DIMENSION, N ODD. METHOD--
   SET THE IMAGINARY PARTS TO ZERO.
4. REAL TRANSFORM FOR THE 1ST DIMENSION, N EVEN. METHOD--
   TRANSFORM A COMPLEX ARRAY OF LENGTH N/2 WHOSE REAL PARTS
   ARE THE EVEN NUMBERED REAL VALUES AND WHOSE IMAGINARY
   PARTS ARE THE ODD NUMBERED REAL VALUES. SEPARATE AND SUP-
   PLY THE SECOND HALF BY CONJUGATE SYMMETRY.

70 ICASE=1
   IFMIN=1
   IIRNG=NPI
   IF(IDIM-4)71,100,100
   IF(IFORM)72,72,100
   ICASE=2
   IIRNG=NPI*(1+NPREV/2)
   IF(IDIM-1)73,73,100
   ICASE=3
   IIRNG=NPI
   IF(NTWO-NPI)100,100,74
   ICASE=4
   IFMIN=2
   NTWO=NTWO/2
   N=N/2

```





MAD07260  
MAD07270  
MAD07280  
MAD07290  
MAD07300  
MAD07310  
MAD07320  
MAD07330  
MAD07340  
MAD07350  
MAD07360  
MAD07370  
MAD07380  
MAD07390  
MAD07400  
MAD07410  
MAD07420  
MAD07430  
MAD07440  
MAD07450  
MAD07460  
MAD07470  
MAD07480  
MAD07490  
MAD07500  
MAD07510  
MAD07520  
MAD07530  
MAD07540  
MAD07550  
MAD07560  
MAD07570  
MAD07580  
MAD07590  
MAD07600  
MAD07610  
MAD07620  
MAD07630  
MAD07640  
MAD07650  
MAD07660  
MAD07670  
MAD07680  
MAD07690  
MAD07700  
MAD07710  
MAD07720  
MAD07730

42C W3R=W2R\*WR-W2I\*WI  
W3I=W2R\*WI+W2I\*WR  
DC 530 I1=1, IIRNG, 2  
KMIN=I1+IPAR\*H  
IF(MMAX-NP1)430,430,440  
430 KMIN=I1  
440 KDIF=IPAR\*MMAX  
45C KSTEP=4\*KDIF  
IF(KSTEP-NT\*G1)460,460,530  
46C DC 520 K1=KMIN,NTCT,KSTEP  
K2=K1+KDIF  
K3=K2+KDIF  
K4=K3+KDIF  
IF(MMAX-NP1)470,470,480  
470 U1R=DATA(K1)+DATA(K2)  
U1I=DATA(K1+1)+DATA(K2+1)  
U2R=DATA(K3)+DATA(K4)  
U2I=DATA(K3+1)+DATA(K4+1)  
U3R=DATA(K1)-DATA(K2)  
U3I=DATA(K1+1)-DATA(K2+1)  
IF( I SIGN)471,472,472  
471 U4R=DATA(K3+1)-DATA(K4+1)  
U4I=DATA(K3)-DATA(K4)  
GO TO 510  
472 U4R=DATA(K4+1)-DATA(K3+1)  
U4I=DATA(K3)-DATA(K4)  
GC TO 510  
480 T2R=W2R\*DATA(K2)-W2I\*DATA(K2+1)  
T2I=W2R\*DATA(K2+1)+W2I\*DATA(K2)  
T3R=WR\*DATA(K3)-WI\*DATA(K3+1)  
T3I=WR\*DATA(K3+1)+WI\*DATA(K3)  
T4R=W3R\*DATA(K4)-W3I\*DATA(K4+1)  
T4I=W3R\*DATA(K4+1)+W3I\*DATA(K4)  
U1R=DATA(K1)+T2R  
U1I=DATA(K1+1)+T2I  
U2R=T3R+T4R  
U2I=T3I+T4I  
U3R=DATA(K1)-T2R  
U3I=DATA(K1+1)-T2I  
IF( I SIGN)49C,500,500  
49C U4R=T3I-T4I  
U4I=T4R-T3R  
GC TO 510  
500 U4R=T4I-T3I  
U4I=T3R-T4R  
51C DATA(K1)=U1R+U2R  
DATA(K1+1)=U1I+U2I  
DATA(K2)=U3R+U4R

MAD07740  
MAD07750  
MAD07760  
MAD07770  
MAD07780  
MAD07790  
MAD07800  
MAD07810  
MAD07820  
MAD07830  
MAD07840  
MAD07850  
MAD07860  
MAD07870  
MAD07880  
MAD07890  
MAD07900  
MAD07910  
MAD07920  
MAD07930  
MAD07940  
MAD07950  
MAD07960  
MAD07970  
MAD07980  
MAD07990  
MAD08000  
MAD08010  
MAD08020  
MAD08030  
MAD08040  
MAD08050  
MAD08060  
MAD08070  
MAD08080  
MAD08090  
MAD08100  
MAD08110  
MAD08120  
MAD08130  
MAD08140  
MAD08150  
MAD08160  
MAD08170  
MAD08180  
MAD08190  
MAD08200  
MAD08210

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520 DATA(K2+1)=U3I+U4I
DATA(K3)=U1R-U2R
DATA(K3+1)=U1I-U2I
DATA(K4)=U3R-U4R
DATA(K4+1)=U3I-U4I
KPIF=KSTEP
KMIN=4*(KMIN-1I)+1I
GO TO 450
530 CCNTINUE
M=M+LMAX
IF(M-MMAX)540,540,570
540 IF(I SIGN)550,560,560
550 TEMPR=WR
WR=(WR+WI)*RTHLF
WI=(WI-TEMPR)*RTHLF
GO TO 410
560 TEMPR=WR
WR=(WR-WI)*RTHLF
WI=(TEMPR+WII)*RTHLF
GO TO 410
570 CCNTINUE
IPAR=3-IPAK
MMAX=MMAX+HMAX
GO TO 360

C MAIN LOOP FOR FACTORS NOT EQUAL TO TWO. APPLY THE TWIDDLE FAC-
C TOR W=EXP(I SIGN*2*PI*SQRT(-1))*(J1-1)*(J2-J1)/(IFP1+IFP2)),
C THEN PERFORM A FOURIER TRANSFORM OF LENGTH IFACT(IF), MAKING USE
C OF CONJUGATE SYMMETRIES.
600 IF(NTWO-NP2)605,700,700
605 IFP1=NTWC
IF=INDN2
NPIHF=NPI/2
IFP2=IFACT(IF)*IFP1
JIMIN=NPI+1
IF(JIMIN-IFP1)615,615,640
615 DO 635 J1=JIMIN,IFP1,NPI
THETA=-TWCP1*FLOAT(J1-1)/FLOAT(IFP2)
IF(I SIGN)625,620,620
THETA=-THETA
WSTPR=COS(THETA)
WSTPI=SIN(THETA)
WR=WSTPR
WI=WSTPI
J2MIN=J1+IFP1
J2MAX=J1+IFP2-IFP1
DO 635 J2=J2MIN,J2MAX,IFP1

```

MAD08220  
MAD08230  
MAD08240  
MAD08250  
MAD08260  
MAD08270  
MAD08280  
MAD08290  
MAD08300  
MAD08310  
MAD08320  
MAD08330  
MAD08340  
MAD08350  
MAD08360  
MAD08370  
MAD08380  
MAD08390  
MAD08400  
MAD08410  
MAD08420  
MAD08430  
MAD08440  
MAD08450  
MAD08460  
MAD08470  
MAD08480  
MAD08490  
MAD08500  
MAD08510  
MAD08520  
MAD08530  
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MAD08570  
MAD08580  
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MAD08630  
MAD08640  
MAD08650  
MAD08660  
MAD08670  
MAD08680  
MAD08690

```

IIMAX=J2+IIRNG-2
DO 630 J1=J2,IIMAX,2
TEMPR=DATA(J3)
DATA(J31)=DATA(J3)*WR-DATA(J3+1)*WI
DATA(J3+1)=TEMPR*WI+DATA(J3+1)*WR
TEMPR=WR
WR=WR*WSTPR-WI*WSTPI
WI=TEMPR*WSTPI+WI*WSTPR
THETA=-TWCPJ/FLOAI(IFACT(IF))
IF(IISIGN)650,645,645
THETA=-THETA
WSTPR=COS(THETA)
WSTPI=SIN(THETA)
J2RNG=IFP1*(1+IFACT(IF)/2)
DO 695 I1=1,IIRNG,2
DO 695 I3=1,IINTOF,NP2
J2MAX=I3+J2RNG-IFP1
DO 690 J2=I2,J2MAX,IFP1
J1MAX=J2+IFP1-NP1
DO 680 J1=J2,J1MAX,NP1
J3MAX=J1+NP2-IFP2
DO 680 J3=J1,J3MAX,IFP2
JMIN=J3-J2+I3
JMAX=JMIN+IFP2-IFP1
I=1+(J3-I3)/NPIHF
IF(J2-I3)655,655,665
SUMR=0.
SUMI=0.
GO 660 J=JMIN,JMAX,IFP1
SUMR=SUMR+DATA(J)
SUMI=SUMI+DATA(J+I)
WORK(I)=SUMR
WORK(I+1)=SUMI
GO TO 680
ICONJ=1+(IFP2-2*J2+I3+J3)/NPIHF
J=JMAX
SUMR=DATA(J)
SUMI=DATA(J+I)
OLDSR=0.
J=J-IFP1
TEMPR=SUMR
SUMR=TWOWR*SUMR-OLDSR+DATA(J)
TEMPI=SUMI
SUMI=TWOWR*SUMI-OLDSI+DATA(J+1)
OLDSR=TEMPR
OLDSI=TEMPI

```

63C

635

640

645

650

655

66C

66E

67C

MAD08700  
MAD08710  
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MAD09080  
MAD09090  
MAD09100  
MAD09110  
MAD09120  
MAD09130  
MAD09140  
MAD09150  
MAD09160  
MAD09170

```

J=J-IFP1
IF(J-JMIN)675,675,670
TEMPR=WR*SUMR-OLDSR+DATA(J)
TEMPI=WI*SUMI
WORK(I)=TEMPR-TEMPI
WORK(ICONJ)=TEMPR+TEMPI
TEMPR=WR*SUMI-OLDSI+DATA(J+1)
TEMPI=WI*SUMR
WORK(I+1)=TEMPR+TEMPI
WORK(ICONJ+1)=TEMPR-TEMPI
CONTINUE
IF(J2-I3)685,685,686
WR=WSIPR
WI=WSIPI
GO TO 690
TEMPR=WR
WI=WSIPR*WSIPI+WI*WSTPR
TWOWR=WR*WR
I=I+1
I2MAX=I3+NP2-NP1
GO 695 I2=I2, I2MAX, NP1
DATA(I2)=WORK(I)
DATA(I2+1)=WORK(I+1)
I=I+2
IF=IF+1
IFP1=IFP2
IF(IFP1-NP2)610,700,700
COMPLETE A REAL TRANSFORM IN THE 1ST DIMENSION, N EVEN, BY CON-
JUGATE SYMMETRIES.
GO TO (900,800,900,701), ICASE
NHALF=N
N=N+N
THETA=-TWCP1/FLOAT(N)
IF(ISIGN)701,702,702
THETA=-THETA
WSTPR=COS(THETA)
WSTPI=SIN(THETA)
WR=WSIPR
WI=WSIPI
IMIN=3
JMIN=2*NHALF-1
GO TO 725
J=JMIN
DO 72C I=IMIN,NIOT,NP2
SUMR=(DATA(I)+DATA(J))/2.

```

675

680

685

686

690

695

C  
C  
C  
C

700  
701

702  
703

710

MAD09180  
MAD09190  
MAD09200  
MAD09210  
MAD09220  
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MAD09270  
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MAD09380  
MAD09390  
MAD09400  
MAD09410  
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MAD09560  
MAD09570  
MAD09580  
MAD09590  
MAD09600  
MAD09610  
MAD09620  
MAD09630  
MAD09640  
MAD09650

```

SUMI=(DATA(I+1)+DATA(J+1))/2.
DIFR=(DATA(I)-DATA(J))/2.
DIFI=(DATA(I+1)-DATA(J+1))/2.
TEMPR=WR*SUMI+WI*DIFR
TEMPI=WI*SUMI-WR*DIFR
DATA(I)=SUMR+TEMPR
DATA(I+1)=DIFI+TEMPI
DATA(J)=SUMR-TEMPR
DATA(J+1)=-DIFI+TEMPI
J=J+NP2
IMIN=IMIN+2
JMIN=JMIN-2
TEMPR=WR
WR=WR*WSTPR-WI*WSTPI
WI=TEMPR*WSTPI+WI*WSTPR
IF(IMIN-JMIN)730,73C,740
IF(ABSIGN)731,740,740
DO 735 I=IMIN,NTOT,NP2
DATA(I+1)=-DATA(I+1)
NP2=NP2+NP2
NTOT=NTOT+NTOT
J=NTOT+1
IMAX=NTOT/2+1
IMIN=IMAX-2*NHAF
I=IMIN
GO TO 755
DATA(J)=DATA(I)
DATA(J+1)=-DATA(I+1)
I=I+2
J=J-2
IF(I-IMAX)750,760,760
DATA(J)=DATA(IMIN)-DATA(IMIN+1)
DATA(J+1)=0.
IF(I-J)77C,780,78C
DATA(J)=DATA(I)
I=I-2
J=J-2
IF(I-IMIN)775,775,765
DATA(J)=DATA(IMIN)+DATA(IMIN+1)
DATA(J+1)=0.
IMAX=IMIN
GO TO 745
DATA(1)=DATA(1)+DATA(2)
DATA(2)=0.
GC TO 900

```

72C  
735  
73C  
731  
735  
74C  
745  
75C  
755  
760  
765  
77C  
775  
78C

COMPLETE A REAL TRANSFORM FOR THE 2ND OR 3RD DIMENSION BY  
CCNJUGATE SYMMETRIES.  
C  
C  
C

MAD09660  
MAD09670  
MAD09680  
MAD09690  
MAD09700  
MAD09710  
MAD09720  
MAD09730  
MAD09740  
MAD09750  
MAD09760  
MAD09770  
MAD09780  
MAD09790  
MAD09800  
MAD09810  
MAD09820  
MAD09830  
MAD09840  
MAD09850  
MAD09860  
MAD09870  
MAD09880  
MAD09890  
MAD09900  
MAD09910  
MAD09920  
MAD09930  
MAD09940  
MAD09950  
MAD09960  
MAD09970  
MAD09980  
MAD09990  
MAD10000  
MAD10010  
MAD10020  
MAD10030  
MAD10040  
MAD10050  
MAD10060  
MAD10070  
MAD10080  
MAD10090  
MAD10100  
MAD10110  
MAD10120  
MAD10130

```

C 800 IF(I1RNG-NP1)805,500,900
C 805 DO 860 I3=1,NTOT, NP2
      I2MAX=I3+NP2-NP1
      DC 860 I2=I2, I2MAX, NP1
      IMIN=I2+IIRAG
      IMAX=I2+NP1-2
      JMAX=2*I3+NP1-IMIN
      IF(I2-I3)820,820,810
      JMAX=JMAX+NP2
      IF(IDIM-2)850,850,830
      J=JMAX*NP0
      DO 840 I=IMIN, IMA X, 2
      DATA(I)=DATA(J)
      DATA(I+1)=-DATA(J+1)
      J=J-2
      J=JMAX
      DO 860 I=IMIN, IMA X, NP0
      DATA(I)=DATA(J)
      DATA(I+1)=-DATA(J+1)
      J=J-NP0
C      END CF LOOP ON EACH DIMENSION
C
C 500 NFO=NP1
C 910 NP1=NP2
C 520 NPREV=N
      RETURN
      END
      SUBROUTINE CHBFT
PURPOSE:
SUBROUTINE CHBFT EVALUATES THE COEFFICIENTS OF AN MTH ORDER
POLYNOMIAL P(X)=A(1)+A(2)*X+A(3)*X**2+...+A(M+1)*X**M SUCH
THAT THE MAXIMUM ERROR ABS(P(X(I))-Y(I)) IS A MINIMUM OVER
THE N (N.GT.M+1) SAMPLE POINTS X(I), Y(I), I=1,2,...,N.
X(I) MUST FORM A STRICTLY MONOTONIC SEQUENCE: I.E.
X(1).LT.X(2).LT...X(N). THIS SUBROUTINE IS A CHEBYSHEV
CONVERSION FROM ALGOL TO FORTRAN OF ALGCRITHM 318, NUMBER 12,
CURVE-FIT FROM COMMUNICATIONS OF THE ACM VOL.10, NUMBER 12,
DECEMBER, 1967. THE AUTHOR OF THE ALGOL VERSION WAS
J. BOOTHROYD FROM THE UNIVERSITY OF TASMANIA.
USAGE:
CALL CHEFT (X,Y,N,A,M,RX,RH,R)
DESCRIPTION OF PARAMETERS:
X -ARRAY OF ABSCESSAE DIMENSIONED REAL*4 X(N)

```



MAD10620  
MAD10630  
MAD10640  
MAD10650  
MAD10660  
MAD10670  
MAD10680  
MAD10690  
MAD10700  
MAD10710  
MAD10720  
MAD10730  
MAD10740  
MAD10750  
MAD10760  
MAD10770  
MAD10780  
MAD10790  
MAD10800  
MAD10810  
MAD10820  
MAD10830  
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MAD10860  
MAD10870  
MAD10880  
MAD10890  
MAD10900  
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MAD10980  
MAD10990  
MAD11000  
MAD11010  
MAD11020  
MAD11030  
MAD11040  
MAD11050  
MAD11060  
MAD11070  
MAD11080  
MAD11090

```

DC 4 J=1,MPLUS1
I1=MPLUS2
A(I1)=A(I1)
RHI1=RH(I1)
I=MPLUS1
5 DENCM=RX(I1)-RX(I-J+1)
AI=A(I)
RHI=RH(I)
A(I1)=(A(I1-AI)/DENOM
RHI1)=(RHI1-RHI)/DENOM
I1=I
AI1=AI
RHI1=RHI
I=I-1
IF(I-J) 4,5,5
4 CONTINUE
EQUATE (M+1) THE DIFFERENCE TO ZERO TO DETERMINE H
H=-A(MPLUS2)/RH(MPLUS2)
WITH H KNOWN, COMBINE THE FUNCTION AND DEVIATION DIFFERENCES
DC 6 I=1,MPLUS2
6 A(I)=A(I)+RF(I)*H
COMPUTE POLYNOMIAL COEFFICIENTS
J=M
7 XJ=RX(J)
I=J
AI=A(I)
JPLUS1=J+1
DO 8 I1=JPLUS1,MPLUS1
AI1=A(I1)
A(I1)=AI-XJ#AI1
AI=AI1
I=I1
8 J=J-1
IF(J-1) 9,7,7
9 CONTINUE
REFERENCE DEVIATION IS NOT INCREASING MONOTONICALLY
IF THE REFERENCE DEVIATION IS NOT INCREASING MONOTONICALLY
THEN EXIT
HMAX=ABS(H)
IF(HMAX.GT.PREVH) GO TO 29
A(MPLUS2)=-HMAX
RETURN
29 FIND THE INDEX, IMAX, AND VALUE, HIMAX, OF THE LARGEST ABSOLUTE
ERROR FOR ALL SAMPLE POINTS
A(MPLUS2)=HMAX
PREVH=HMAX
IMAX=P(I)
HIMAX=H
J=1

```

```

MADI11100
MADI11110
MADI11120
MADI11130
MADI11140
MADI11150
MADI11160
MADI11170
MADI11180
MADI11190
MADI11200
MADI11210
MADI11220
MADI11230
MADI11240
MADI11250
MADI11260
MADI11270
MADI11280
MADI11290
MADI11300
MADI11310
MADI11320
MADI11330
MADI11340
MADI11350
MADI11360
MADI11370
MADI11380
MADI11390
MADI11400
MADI11410
MADI11420
MADI11430
MADI11440
MADI11450
MADI11460
MADI11470
MADI11480
MADI11490
MADI11500
MADI11510
MADI11520
MADI11530
MADI11540
MADI11550
MADI11560
MADI11570

12 K=K-1
112 HI=HI*XI+A(K)
IF(K-1) 112, 12, 12
HI=Y(I)
ABSHI=ABS(HI)
IF(ABSHI.LE.HMAX) GO TO 11
HMAX=ABSHI
HMAX=HI
GC TO 110
11 IF(J.GE.MPLUS2) GC TO 110
J=J+1
RJ=R(J)
110 CONTINUE
IF THE MAXIMUM ERRCR OCCURS AT A NONREFERENCE POINT, EXCHANGE THIS
POINT WITH THE NEAREST REFERENCE POINT HAVING AN ERRCR OF THE
SAME SIGN AND REPEAT
IF(IMAX.EQ.R(I)) RETURN
DO 14 I=2,MPLUS2
IF(IMAX.LT.R(I)) GO TO 15
14 CONTINUE
I=MPLUS2
I=MPLUS2
15 NEXTI=H
IF((I-I/2)*NE.0) NEXTI=-H
IF(HIMAX#NEXTI.GE.0) GO TO 115
IF(IMAX.GE.R(I)) GC TO 116
J=M
117 R(J)=R(J)
JI=J
J=J-1
IF(J-1) 118, 117, 117
118 R(I)=IMAX
GO TO 2
116 IF(IMAX.LE.R(MPLUS2)) GO TO 120
J=1
DO 121 JI=1,MPLUS2
R(J)=R(JI)
121 R(MPLUS2)=IMAX
GC TO 2
115 R(I)=IMAX

```

C  
C  
C

MAD11580  
MAD11590  
MAD11600  
MAD11610  
MAD11620  
MAD11630  
MAD11640  
MAD11650  
MAD11660  
MAD11670  
MAD11680  
MAD11690  
MAD11700  
MAD11710  
MAD11720  
MAD11730  
MAD11740

```

12C R(1-1)=IMAX
GO TO 2
GO TO 2
END
//GO.SYSIN DD *
LA MESA VILLAGE, 18 AUG 82 0507-0636 LOCAL.
PSD CF X COIL, AMP IN DB REF NT**2/HZ
LA MESA VILLAGE, 18 AUG 82 0507-0636 LOCAL.
PSD CF Y COIL, AMP IN DB REF NT**2/HZ
LA MESA VILLAGE, 18 AUG 82 0507-0636 LOCAL.
PSD OF Z COIL, AMP IN DB REF NT**2/HZ
LA MESA VILLAGE, 18 AUG 82 0507-0636 LOCAL.
PSD CF TOTAL FIELD, AMP IN DB REF NT**2/HZ
//GO.FT20FOO1 DD UNIT=3400-4, VOL=SER=GMDT4A, DISP=(CLD,KEEP),
// LABEL=(1,1,IN)
// DCB=(RECFM=FB, LRECL=32, BLKSIZE=512, DEN=2)
//GO.SYSOUMP DD SYSOUT=A

```

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