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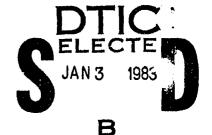
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TECHNICAL REPORT ARLCD-TR-82013

COMPUTER SIMULATION OF ARTILLERY S&A MECHANISM (INVOLUTE GEAR TRAIN AND STRAIGHT-SIDED VERGE RUNAWAY ESCAPEMENT)

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NOVEMBER 1982



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US ARMY ARMAMENT RESEARCH AND DEVELOPMENT COMMAND
LARGE CALIBER
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the two configurations of a complete artillery	safing and arming mechanism that
is driven by a rotor and contains a straight-si	ded verge runaway escapement and
a two-pass involute gear train. The mathematic	al model treats three motion
regimes of the associated escapement; i.e., cou	pled motion, free motion, and
impact. The computer program is well adapted t	o parametric studies, and it

allows the use of pallets with arbitrarily located centers of mass. Further,

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it makes it possible to determine the maximum value of the contact forces in the gear train as well as in the escapement. A simulation run with the dimensions of the M739 safing and arming mechanism showed excellent agreement with experimental results when a system coefficient of friction of 0.1 was used.

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INTRODUCTION

A computer simulation was developed for a complete artillery safing and arming (S&A) mechanism containing a straight-sided verge runaway escapement. Figures 1 and 2 show the essentials of two different configurations of such a mechanism which, in addition to the escapement, contains a spin driven rotor and a two-pass involute gear train.

While the mathematical model of the verge runaway escapement represents a new study, the description of the rotor and the gear train, together with the essence of the computer program, is based to a considerable extent on reference 1, which deals with the computer simulation of an S&A mechanism containing a pin pallet runaway escapement.

As in reference 1, the three motion regimes of the runaway escapement (coupled motion, free motion, and impact) are considered. Again, the pallet of the escapement may have an arbitrarily-located center of mass.

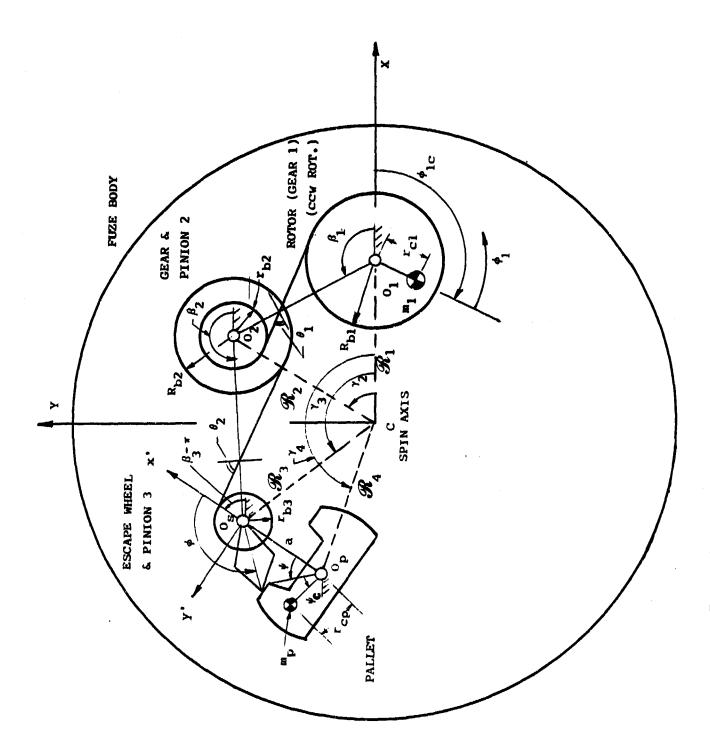
Complete derivations are given for the mathematical model, including the configuration variations. A detailed description and operational instructions are given for the associated computer program whose output furnishes instantaneous kinematic conditions of the escapement components and the number of turns to arm, as well as all non-impact contact forces. This program was extensively tested for the configuration no. 2 data (fig. 2) of the M739 fuze S&A mechanism. This mechanism simulation required approximately 26 turns to arm when a system coefficient of friction of 0.1 was used.

DESCRIPTION OF COMPUTER PROGRAM

The following gives the essential steps of program SANDA2V which is listed in appendix A. The mathematical model of the verge runaway escapement naturally differs from that of the pin pallet escapement as shown in reference 1, especially since the differential equations for entrance- and exit-coupled motions in the verge escapement are not the same. However, the programming schemes are essentially identical. Thus, with the exception of controls pertaining to the aforementioned entrance- and exit-coupled motions, the program runs parallel to the one associated with the pin pallet escapment.

The following outlines the essential features of the program without giving specific control details.

The main program starts with reading in and writing of all relevant physical data. This is followed by the computation of gear ratios, fuze body angles, centrifugal forces, gear train constants, and earliest and latest possible values of the gear angles by way of subroutine ALFA, as well as the initialization of the gear angles. The simulation begins with the entrance-coupled motion at a starting angle PHID, which represents that angle ϕ of the escape wheel which is associated with the approximate center of the entrance working surface of the



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Figure 1. Rotor driven S&A device, configuration no. 1

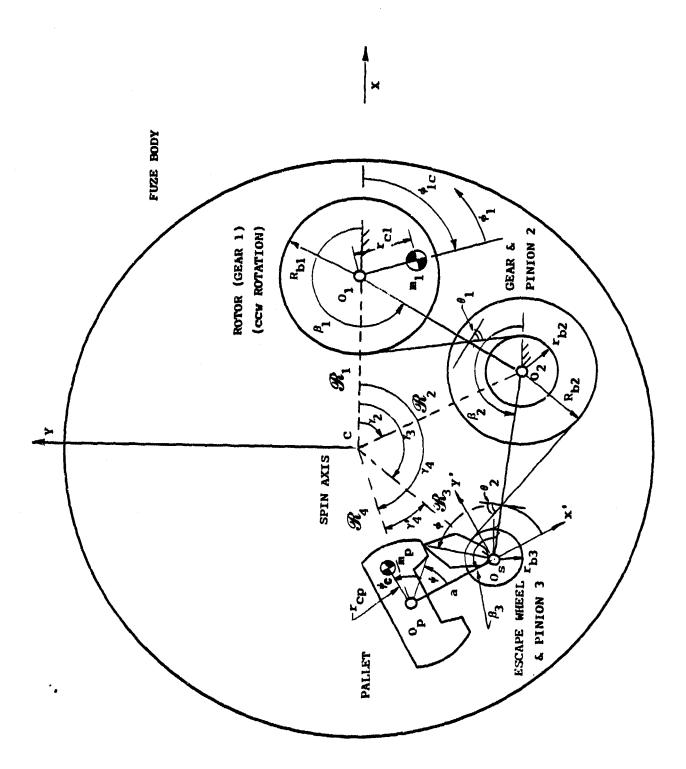


Figure 2. Rotor driven S&A device, configuration no. 2

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pallet. This angle then corresponds to a cumulative escape wheel angle PHITOT of 0° .

Coupled Motion (Location 1)

Depending on whether entrance- or exit-coupled motion takes place, differential equation B-100 or equation B-119 of appendix B must be solved. To this end, the main program calls on an available fourth-order Runge-Kutta routine. The main purpose of the subroutine FCT is to present the second-order differential equation in terms of two first-order ones to RKGS. PHI(1) and PHI(2) represent the angle ϕ and the angular velocity ϕ , respectively. The computation of all parameters of the differential equation takes place in subroutine FCT as well as in subroutines IN2 and KINEM.

The subroutine KINEM computes current values of g, ψ and $\dot{\psi}$, as shown in appendix C, as well as of the moment arms A_1^i , B_1^i , C_1^i , and D_1^i as shown in appendix D.

Subroutine IN2 computes various gear mesh parameters and instantaneous mesh contact angles, as well as the signum functions \mathbf{s}_1 , \mathbf{s}_2 , \mathbf{s}_4 , and \mathbf{s}_5 . In addition, the parameters \mathbf{A}_1 to \mathbf{A}_{44} , listed in appendix B, are obtained. It is to be noted that the parameters \mathbf{A}_1 , \mathbf{A}_5 , \mathbf{A}_9 , \mathbf{A}_{12} , \mathbf{A}_{17} , and \mathbf{A}_{18} depend on whether entrance— or exit—coupled motion is involved. The parameters \mathbf{A}_{45} to \mathbf{A}_{50} are computed in subroutine FCT. The reason for this division is found in the limited number of arguments permitted in any one subroutine. Finally, the gear—indexing operation, as outlined in reference 2, is performed with the help of the angle ϕ .

The following discusses the manner in which the instantaneous rotor angle $\phi_{1C}+N_{31}\phi_{T}$ of the coupled motion differential equations must be expressed in subroutine FCT. Recall that ϕ_{1C} is the initial rotor angle, ϕ_{T} is the total angle of rotation of the escape wheel, and N_{31} stands for the gear ratio between the rotor and the escape wheel. Since the angle ϕ with the Runge-Kutta variable PHI(1) varies between approximately 134° and 144° during entrance— and between 209° and 216° during exit-coupled motion, the total escape wheel angle ϕ_{T} can only be obtained by continuously adding the increments due to each cycle of Runge-Kutta computations. Thus,

$$\phi_{\rm T} = \phi_{\rm TOT} + \Delta \phi \tag{1}$$

RKGS Routine, IBM System/360 Scientific Subroutine Package (360A-CM-0X3), Version III.

² The program uses the symbols AA1, etc. throughout. This should not be confused with the symbols AA₁ to AA₁₈, which are used in the combined exit-coupled motion differential equation (B-119).

where

φ_{TOT} = total escape wheel angle up to a certain Runge-Kutta cycle. (This is represented by PHITOT in the program.)

 $\Delta \phi$ = increment of escape wheel during this Runge-Kutta cycle.

The increment $\Delta \phi$ is calculated as the difference between the latest value of PHI(1) and its previous one which has been stored as PHIPR. In this manner, equation 1 becomes

$$\phi_{\mathbf{r}} = PHITOT + PHI(1) - PHIPR$$
 (2)

Subroutine FCT also decides on the values of I_{pR} and I_{1R} as required by equations B73 and B74 as well as equations B150 and B151 (app B). The associated conditional statements assign the larger values for these combined moments of inertia whenever the product of the angular velocity and the angular acceleration is positive; i.e., both quantities have the same sign.

The associated suboutine OUTP is responsible for printing out the results $\dot{\phi}$, $\dot{\phi}$, and $\dot{\phi}$, together with the current values of time, g, ψ , $\dot{\psi}$, and PHITOT. Further, all coupled motion contact forces are calculated according to equations B183 and B186, and the maximum values of these forces during one-rotor cycle are determined.

Free Motion (Location 5)

The differential equations of free motion, as given by equation B187 for the pallet and equation B193 for the combined escape wheel, gear train, and rotor are again solved by the Runge-Kutta routine. To obtain the magnitudes of the variables φ and ψ , as well as their derivatives at identical times, the two independent second-order differential equations are transformed into four simultaneous first-order ones. (While only the two first-order equations associated with each of the two variables are actually coupled, the routine treats all four as if they were coupled and thus produces solutions for identical time increments.) These four expressions, which are presented in subroutine FCTF, are of the following form:

$$DX(1) = X(2) : (= \dot{\phi})$$
 (3)

$$DX(3) = X(4) : (= \psi)$$
 (4)

$$DX(2) = \frac{1}{A_{55}} \left[-A_{56}(X(2))^2 + A_{57}X(2) + A_{58} + A_{59}\sin(\phi_{1C} + N_{31}(\phi_{T} + X(1) - PHIPR)) \right] : (=\phi)$$
(5)

$$DX(4) = \frac{1}{A_{51}} \left[-A_{21} (X(4))^{2} - A_{52}X(4) - A_{53} + A_{54}\sin(\gamma_{P}^{*} - \psi - \psi_{C}) \right] : (=\psi)$$
(6)

The subroutine FCTF also computes the parameters A_{51} to A_{59} and calls on subroutine IN2 for the computation of all gear-related parameters.

The associated subroutine OUTPF computes the free motion contact forces according to equations B199 and B200 and finds their maxima after determining the parameters A_{51} to A_{59} . The computation of these latter variables also requires a call on subroutine IN2. In addition to the above, a continuous count of PHITOT is provided in OUTPF. This angle, time, ϕ , ϕ , ψ , ψ , and the contact forces are printed out. This subroutine also makes the decision whether or not to remain in free motion. The sensing variables f and g' = GP (equations E4 and E5 of appendix E) are used for this purpose.

Impact (Location 15)

The subroutine IMPACT uses the current values of the angular velocities $\dot{\phi}_1$ and $\dot{\psi}_1$ and computes the post impact angular velocities $\dot{\phi}_f$ and $\dot{\psi}_f$, applying equations B211 and B212 of appendix B. (Note that the moment of inertia of the escape wheel is now expressed according to equation B213 (app B), which refers the rotor as well as the gear train inertias to the escape wheel.)

Reversal of Gear Train Motion Due to Impact

If the impact torque on the escape wheel is sufficiently large, the motion of the gear train may be temporarily reversed; i.e., the escape wheel angular velocity $\dot{\phi}$ may become negative. This would cause the friction forces between the gear teeth and at the various gear pivots to be reversed in direction. (The normal forces between the gear teeth remain unaffected, and the normal bearing forces are obtained in the usual manner.) This change in the direction of the friction forces is expressed for both coupled and free motion by letting the coefficient of friction μ of all gear train components become negative (discussed in app E of ref 1). This is accomplished in subroutine IN2 by the following use of the signum function $\dot{\phi}/|\dot{\phi}|$:

$$MU = ABS(MU) * \dot{\phi}/|\dot{\phi}| \tag{7}$$

Whenever $A_{51} = I_{PR} = 0$, the simulation stops because of the division by zero. Should this occur, FCTF prints "IPR EQUALS ZERO - SIMULATION TERMINATED."

(The coefficient of friction associated with the escapement interface and the pallet pivot is called ν_1 and is read into the program as MUI.) Any motion reversal at these surfaces is accounted for by the signum functions s₄ and s₅, respectively.

Termination of Computations

Computations are terminated whenever the geared motion of the rotor ends. This corresponds to ϕ = PHICUTD. The duration of the subsequent unretarded motion of the rotor is assumed to be negligible.

COMPUTER SIMULATION OF EXAMPLE MECHANISM

This section discusses a computer simulation of a modified S&A device of the M739 fuze. The mechanism has configuration no. 2, as shown in figure 2, and contains a newly-designed involute gear train. While this gear train has the same gear ratio and individual center distances as the original design, each of the meshes now has unity contact ratio. The simulation of this mechanism was accomplished with the help of computer program SANDA2V which, with a sample output, is listed in appendix A. It was run for 30,000 rpm to obtain maximum contact forces.

The following shows the input requirements of the program, explains the various output data, and discusses the manner in which the "number-of-turns-to-arm" is obtained for a given spin velocity.

Input Data (Appendix A)

The first portion of the output repeats all input data, which represent the mechanism parameters of the M739 fuze. These are listed both as computer variables and as symbols, according to appendixes C and F of this report:

Both meshes were designed with the help of computer programs INVOL11 and GEARPARAM2, originally shown in Progress Report No. 11 of the "Development of Automated Design Optimization Technique for Safety and Arming Devices" (Contract No. DAAK10-79-C-0251, January 15, 1981). Copies of this report may be obtained from either F. R. Tepper, ARRADCOM or G. G. Lowen, The City College of New York.

Escapement Parameters

A = a = 0.226 in. (5.740 mm) = distance between pivots O_p and O_s (fig. 2)

B = b = 0.168 in. (4.267 mm) = escape wheel radius

C = c = 0.13138 in. (3.337 mm) = pallet radius as defined by figure F-1 of appendix F

ALPHEN = α_{en} = 43.5352° = entrance working surface angle

ALPHEX = α_{ex} = 29.2981° = exit working surface angle

NT = 4 = number of escape wheel teeth spanned by verge

CONFIG = 2 = Configuration no. 2 (See section on fuze body configuration no. 2 in appendix B.)

EREST = $e_r = 0$ = Coefficient of restitution

LAMBDA = λ = 92.93° = angle between entrance and exit pallet radii (figure F-1 in appendix F)

N = 22 = Number of escape wheel teeth

For details of the above nomenclature, see appendixes C, E, and F.

Mass Parameters of Components

$$M1 = m_1 = 0.3165 \times 10^{-4} \text{ lb-sec}^2/\text{in.} (5.552 \times 10^{-3} \text{ kg}) = \text{mass of rotor}$$

$$M2 = m_2 = 0.3275 \times 10^{-5} \text{ lb-sec}^2/\text{in.}$$
 (5.745 x 10^{-4} kg) = mass of gear and pinion no. 2

$$M3 = m_3 = 0.2631 \times 10^{-5} \text{ lb-sec}^2/\text{in.}$$
 (4.615 x 10^{-4} kg) = mass of escape wheel and pinion no. 3

$$MP = m_p = 0.1640 \times 10^{-5} \text{ lb-sec}^2/\text{in.} (2.877 \times 10^{-4} \text{ kg}) = \text{mass of pallet}$$

II =
$$I_1 = 0.1967 \times 10^{-5} \text{ in.-lb-sec}^2 (2.226 \times 10^{-7} \text{ kg-m}^2) = \text{moment of inertia of rotor}$$

$$I2 = I_2 = 0.4026 \times 10^{-7} \text{ in.-lb-sec}^2 (4.556 \times 10^{-9} \text{ kg-m}^2) = \text{moment of inertia of gear and pinion no. 2}$$

I3 =
$$I_3 = 0.2125 \times 10^{-7}$$
 in.-1b-sec² (2.405 x 10^{-9} kg-m²) = moment of inertia of escape wheel and pinion no. 3

IP = $I_p = 0.1950 \times 10^{-7} \text{ in.-lb-sec}^2 (2.207 \times 10^{-9} \text{ kg-m}^2) = \text{moment of inertia of pallet}$

General Parameters

RC1 = r_{c1} = 0.0576 in. (1.463 mm) = distance from pivot of rotor to its center of mass

RCP = r_{cp} = 0. = pallet eccentricity

RHOP = $\rho_{\rm p}$ = 0.0227 in. (0.577 mm) = pallet pivot radius

RPM = 30,000 = spin rate

PHIICD = ϕ_{1c} = -120.134° = rotor angle in starting position (fig. 2)

PSICCD = ψ_c = 0° = eccentricity angle of pallet

PHID = 139° = escape wheel starting angle of initial coupled motion

PHICUTD = 1485° = cumulative escape wheel angle obtained from product of total engaged rotor rotation and gear ratio. The total rotor rotation for the M739 fuze is 46.41°, while the gear ratio is 32. Thus, PHICUTD = 46.41 x 32 = 1485°

MU = μ = 0.10 = coefficient of friction of gear train (pivots and tooth-to-tooth contacts) and escape wheel pivot

MU1 = μ_1 = 0.10 = coefficient of friction of pallet-escape wheel interface and pallet pivot

Gear Parameters

PSUBD1 = P_{d1} = 80 = diametral pitch of mesh no. 1 (rotor and pinion no. 2)

PSUBD2 = P_{d2} = 100 = diametral pitch of mesh no. 2 (gear no. 2 and escape wheel pinion)

 $NG1 = N_{G1} = 64 = number of teeth of rotor (full gear no. 1)$

 $NG2 = N_{G2} = 36 = number of teeth of gear no. 2$

 $NP2 = N_{p2} = 9 = number of teeth of pinion no. 2$

NP3 = N_{P3} = 8 = number of teeth of pinion no. 3 (escape wheel pinion)

```
CAPRP1 = R_{P1} = 0.41214 in. (10.468 mm) = pitch radius of gear no. 1 (rotor)
```

CAPRP2 = R_{p2} = 0.19039 in. (4.836 mm) = pitch radius of gear no. 2

 $RP2 = r_{D2} = 0.05796$ in. (1.472 mm) = pitch radius of pinion no. 2

RP3 = r_{p3} = 0.04231 in. (1.075 mm) = pitch radius of pinion no. 3 (escape wheel pinion)

THETAl = θ_1 = 24.215° = pressure angle of mesh no. 1

THETA2 = θ_2 = 27.326° = pressure angle of mesh no. 2

 $R1 = R_1 = 0.250$ in. (6.350 mm) = distance of rotor pivot from spin axis

 $R2 = R_2 = 0.317$ in. (8.052 mm) = distance of pivot of gear and pinion set no. 2 from spin axis

R3 = \Re_3 = 0.309 in. (7.849 mm) = distance of pivot of escape wheel from spin axis

 $R4 = R_4 = 0.304$ in. (7.722 mm) = distance of pivot of pallet from spin axis

RH01 = ρ_1 = 0.03075 in. (0.781 mm) = pivot radius of rotor

RH02 = ρ_2 = 0.015 in. (0.381 mm) = pivot radius of gear and pinion no. 2

RH03 = ρ_3 = 0.015 in. (0.381 mm) = pivot radius of escape wheel

CAPRB1 = R_{b1} = 0.37588 in. (9.547 mm) = base radius of gear no. 1

CAPRB2 = R_{b2} = 0.16915 in. (4.296 mm) = base radius of gear no. 2

RB2 = r_{b2} = 0.05286 in. (1.343 mm) = base radius of pinion no. 2

RB3 = r_{b3} = 0.03759 in. (0.955 mm) = base radius of escape wheel pinion

CAPRO1 = $R_{0.1}$ = 0.41425 in. (10.522 mm) = outside radius of gear no. 1

 $CAPRO2 = R_{02} = 0.19404$ in. (4.929 mm) = outside radius of gear no. 2

 $R02 = r_{02} = 0.07670$ in. (1.948 mm) = outside radius of pinion no. 2

 $R03 = r_{03} = 0.05580$ in. (1.417 mm) = outside radius of escape wheel pinion

 $J1 = J_1 = 0$ = initialization parameter for mesh no. 1 [The zero value corresponds to earliest possible contact of mesh (ref 3).]

 $J2 = J_2 = 0 = initialization parameter for mesh no. 2$

Output Data

The data blocks following the input data represent the results of various computations.

Fuze Geometry

The angles BETAID = β_1 to BETA3D = β_3 and GAMMA2D = γ_2 to GAMMA4D = γ_4 are printed for checking purposes.

Coupled Motion

The first coupled motion output refers to the entrance side of the verge. For each time T of the coupled motion, the following variables are computed:

PHI = ϕ = instantaneous escape wheel angle (deg)

PHIDOT = ϕ = escape wheel angular velocity (rad/sec)

G = g = pallet - escape wheel contact position (in.) (equation Cl5 of appendix C)

PSID $= \psi = pallet angle (deg)$

PSIDOT = $\dot{\psi}$ = pallet angular velocity (rad/sec)

PHITOT = $\phi_{\mathbf{T}}$ = cumulative escape wheel angle (deg)

 $F23 = F_{23} = normal$ contact force of gear no. 2 on pinion no. 3 (1b)

 $F12 = F_{12} = normal$ contact force of gear no. 1 on pinion no. 2 (1b)

 $PN = P_n = normal$ contact force between escape wheel and pallet (1b), computed according to equation 3185 in appendix B

PNPSI = P = normal contact force between escape wheel and pallet (1b), computed according to equation B186 in appendix B (serves for checking)

DPHI2 = ϕ = escape wheel angular acceleration (rad/sec²), Runge-Kutta output

Free Motion

The first free motion on the exit side follows the coupled motion on the entrance side of the verge. For each time T of the free motion, the following variables are evaluated:

PHI = ϕ = instantaneous escape wheel angle (deg)

PHIDOT = $\dot{\phi}$ = escape wheel angular velocity (rad/sec)

PSI = ψ = pallet angle (deg)

PSIDOT = ψ = pailet angular velocity (rad/sec)

PHITOT = ϕ_T = cumulative escape wheel angle (deg)

 $FF12 = F_{F12} = normal$ contact force of gear no. 1 on pinion no. 2 for free motion (1b)

 $FF23 = F_{F23} = normal contact force of gear no. 2 on escape wheel pinion for free motion (1b)$

Impact

The first exit impact follows the first exit free motion. Just preceding the "IMPACT" label, the program prints the values of VP = $V_{\rm TNi}$ and VS = $V_{\rm SNi}$ which stand for the pre-impact velocity components, normal to the verge face, of both the pallet and escape wheel contact points (equations D13 and D15). Subsequent to the "IMPACT" label, the following variables are evaluated:

PHI = ϕ = instantaneous escape wheel angle (deg), same as before impact

PHIDOT = $\dot{\phi}$ = post-impact escape wheel angular velocity (rad/sec)

PSI = ψ = pallet angle (deg), same as before impact

PSIDOT = $\dot{\psi}$ = post-impact pallet angular velocity (rad/sec)

PHITOT = ψ_{T} = cumulative escape wheel angle (deg), same as before impact

 $VP = V_{TNf}$ = post-impact normal velocity component of pallet at contact point (equation D15)

VS = V_{SNf} = post-impact normal velocity component of escape wheel tooth at contact point (equation D13)

Note that in the present program, the post-impact VP is equal to VS since the coefficient of restitution is zero.

"Number of Turns-to-Arm" and Maximum Contact Forces

The "number of turns-to-arm" at 30,000 rpm is obtained with the help of that time T_{1485} which corresponds to the escape wheel angle PHICUTD = 1485°. Thus,

number of turns-to-arm =
$$\frac{30000}{60}$$
 x 0.05156 = 25.8

The maximum non-impact contact forces for the total cycle, for both coupled and free motion, are listed at the end of the output.

CONCLUSIONS

While it was not the purpose of this investigation to undertake a parametric study of the mechanism for which the program was written, the program was sufficiently tested to confirm that such a study is possible. It may include variations in masses and moments of inertia of all components; variations in the locations of the centers of mass of the verge and the rotor; and variations of gear, escapement, and fuze geometries; as well as various friction and coefficient of restitution conditions.

Previous high-speed motion picture observations of pin pallet escapements showed that the impacts were essentially inelastic and that, therefore, a zero coefficient of restitution was justified. Similar observations made on the detached lever escapement of the M577 fuze timer confirmed this.

The present work reports only on a single test run using the M739 fuze S&A data with a system coefficient of friction of 0.1. This is assumed to be representative of actual test conditions since (1) previous simulations of pin pallet escapements showed that the range of actual experimental results may be reproduced with coefficients of friction between 0.1 and 0.2 and (2) a special lubricant is used in conjunction with the M739 fuze. This choice of coefficient of friction is borne out by the good agreement with experimental results.

Noteworthy is the fact that the mechanism armed in approximately 22 turns with a coefficient of friction of 0.05 and in approximately 31 turns with a coefficient of friction of 0.15.

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APPENDIX A COMPUTER PROGRAM SANDA2V

PROCESSA & SUDDAY (PURITY) TABLE & SUDDAY (PURITY)		FTM 4.8+508	06/08/82	13.52.46	PAGE	-
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8		32, DMEGA, DMZ, RCT, PHI	× ×			
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REAL MI, MZ, M3, MP, T1, 12, 73, TP, LAMBDA, K, M31, M32, JJ, JJZ, M51, M62, M72, M7 SAML, M11, M7, M SAMLE DATA READ (5, 17) A 19, CA, APPEN, ALPHEN, ALPHEN, M1, CONFIG READ (5, 17) A 19, CA, APPEN, ALPHEN, M1, CONFIG READ (5, 17) M1, M2, M3, M9 MRITE (6, 29) REST, LAMBDA, M READ (5, 27) M1, M2, M3, M9 MRITE (6, 29) M1, M2, M3, M3 MRITE (6, 29) M1, M2, M3, M3, M3, M3, M3, M3, M3, M3, M3, M3	2	K2(8.4), PRMT(5), PHI(2).	6 Y			
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READ IN AND WRITE DATA			A 13			
READ (5,17) A, B,C,ALPHEN, ALPHEX, NT, COMFIGENARM	·	READ IN AND WRITE DATA	A 15			
READ (5,19) REST, LAMBOA, N CONFIG	9		A 16			
READ (5,19) REST, LANDA, N		B.C.ALPH	A 17			
RED 6,20 EREST LABOA, N	2	A.B.C.ALP	8 -			
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READ (5.25) RC1, RC2, RRD1, SPECCO, PHID, PHICUTO, MU, MU1		(5.24)	A 25			
READ 15,26 PSUBO1, PSUBO2, NG1, NG2, NP2, NP3, CAPRP2, RP2, RP3, TA		(6,25)	A 26			
WRITE (6.29) PSUBD1, PSUBD2, NG1, NG2, NP2, NP3, CAPRP1, CAPRP2, RP2, TA A HETA1, THETA2 **READ (5.27) R1, R2, R3, R4 **READ (5.19) RHO1, RHO2, RHO3 **READ (5.27) CAPRB1, CAPRB2, RB3 **READ (5.27) CAPRB1, CAPRB2, RB3 **READ (5.28) J1, J2 C INITIALIZATION OF PARAMETERS AND CONVERSION TO RADIANS **PHINE=0.** **PHINE=PHID A PARAMETERS AND CONVERSION TO RADIANS **PHINE A PARAMETERS AND CONVERSION TO RADIANS		(5,26) PSUBO1, PSUBD2, NG1, NG2, NP3, CAPRD1, CAPRD1, BBD, TA	A 27			
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MORE MORE	06/08/82 13.52.46 PAGE 2	929	1 60	1 62 1 64	99 1	68	1 72	1 75 1 75 1 76	1 79	81	1 84 1 85	1 87 1 88	1 90 1 91	1 93 1 94	96 1 97	1 99 1 100	102	105	108	110
		DM2=DMEGA+ONEGA PHI1C=PHI1CD+22	PSICC=PSICCD*22 PSIC=PSICC At Phen at Dien*27		COMPUTATION OF	N31=NP2+NP3/(NG1+NG2) N32=-NP3/NG2	IF (CONFIG.EQ.1.) SG=1	COMPUTATION OF GAMMAS	GAMMA2= S6+ACOS ((R1+R1+ GAMA3P = ACOS ((R2+R2+R3+ GAMMA3 = GAMMA2+S6+GAMA3	GAMMA4P=ACOS((R3+R3+R4-A+A)/(2.+R3+R4)) GAMMA4=GAMMA3+S6+GAMA4P GAMMA2D=GAMMA2/72	GANMA3D=GANMA3/2Z GANMA4D=GANMA4/ZZ	DELIA3=ACOS(((CAPR2+R3-R3-R2-R3-R3)/(2.*R2*(CAPRP2+RP3))) DELIA3=ACOS((A**H3*R3-R4*R4)/(2.*A*R3))	BETA2=GAMA2+PI-S6*DELTA3 BETA3=GAMA3+PI-S6*DELTA4	SARA .	TA10,BE	CONVERSION OF PRESSURE ANGLES TO	THETA2=THETA2+22	T2=M2+R2+OM2	DETERMINATION OF GEAR	TEST1=TAN(THETA1) TEST2=TAN(THETA1) TEST2=TAN(THETA1) D2=(CAPRB2+RB3)+TAN(THETA1)

115	4 OPT=1 ION OF EARL!	06/08/82 A 115 A 116 A 117	13.52.46	PAGE	•
120	ALEA (CAPRB2, RB3) ALIZATION OF ALP	A 119 A 120 A 121 A 121			
*	C DATA FOR RUNGE KUTTA C ALPHR-ALPHEN PRMT(2)=3.	A 123 A 125 A 125 A 125 A 127 A 127			
8 8]-].	A 129 A 131 A 132 A 132 A 133			
94		A 136 A 137 A 138 A 140 A 140			
145	2 CALL RKGS (PRMT, PHI, DPHI, NDIM, IHLF, FCT, OUTP, AUX) 1	A A A A A A A A A A A A A A A A A A A			
8	IF (PHID.GE.130.00.AND.PHID.LE.160.) GO TO 3 GO TO 4 3 PHI(1)=PHI(1)+DELTA*22*NT PHIPR=PHI(1)/ZZ PSIC=PSICZ+LAMBDA*ZZ ALPHR=ALPHEX	A 150 A 153 A 153 A 153 A 154 A 155 A 155			
160	GD TO 9 4 PHI(1)=PHI(1)-DELTA*ZZ*(NT+1.) PHIPR=PHI(1)/ZZ PSI=PSI-2.*PI+LAMBDA*ZZ ALPHR=ALPHEN PSIC=PSICC	A 158 A 159 A 160 A 161 A 161			
201	C FREE MOTION C X(1)=PHI(1) X(2)=PHI(2) X(3)=PSI X(4)=DPSI	A 163 A 164 A 165 A 166 A 169 A 170 A 171			

FTN 4.8+508

C 11 IF [PH1(2).GE. D. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 E IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). CG. ABS(VS)) GO 10 12 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). CG. ABS(VS)) GO 10 12 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). LT. ABS(VS)) GO 10 12 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). LT. ABS(VS)) GO 10 12 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). LT. ABS(VS)) GO 10 12 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). LT. ABS(VS)) GO 10 12 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). LT. ABS(VS)) GO 10 14 PHIDPAPHIC (PH1(1).ZE. O. AND. DPS1.GE. O. AND. ABS(VP). LT. ABS(VS)) GO 10 14 C EXIT ACTION C EXIT ACTION C EXIT ACTION F (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP). GT. ABS(VS)) GO 10 11 IF (PH1(2).GE. O. AND. DPS1.GE. O. AND. ABS(VP	238		20/00/00	2	•
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F (PHI (2) (16. 0. AND D951, GE 0. AND ABS(PP), LT, ABS(VS) GD TG 12 F (PHI (2) (16. 0. AND D951, GE 0.) GD TG 12 F (PHI (2) (16. 0. AND D951, GE 0.) GD TG 12 F (PHI (2) (16. 0. AND D951, GE 0.) GD TG 12 F (PHI (2) (16. 0. AND D951, GE 0.) AND ABS(VP), GT, ABS(VS) GD TG 12 F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) GD TG 12 F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) GD TG 1 F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) GD TG 1 F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) GD TG 1 F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, LE 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951, GE, 0. AND ABS(VP), GT, ABS(VS) F (PHI (2) (16. 0. AND D951	238	(PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).EG.ABS(VS)) GO TO	A 231		
F (FM12).LE. 0.1 AND LOSSI.LE. 0.1 AND LASS(VP). CO. ABS(VS); GD TO 12	238	(PHI (2). UE. 0 AND. UPSI.GE. 0 AND. ABS(VP). LT. ABS(VS)) GO TO	A 232		
F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . AND AND A8S(VP) LT A8S(VS) GD TO 1 F (PHI(2) LE 0 . AND D951 LE 0 . A	23	CONTRACTOR OF SUB-DEST OF STATES	A 233		1
		(***(*).GE-0**NO-04-51.EE-0) UG [U [2] {But(9) ! f 0 Ann f8ct f 0 Ann Antwork : + Andrews on	A 234		
Teach Teac		(PHI(2) IE.O. AND DPSI IE.O. AND ARCIVE) OF ARCIVE) CO TO	A 235		
C TEST FOR EXIT ACTION PHID-PHI(1)/ZZ FMID-PHI(1)/ZZ FMID-PHI(2)/ZZ FMID-PHICA/ZZ FMID-PHICA/		(PHI(2), 1E.O. AND, DPST, 1E.O. AND ARC(VD) ED ARC(VC) OF TO	A 233		
C CALL IMPACT (FHI(1) PMI(2), PSI, DPSI) C TEST FOR EXIT ACTION PHID-PHI(11/2Z) IF (FMID.LE.160.0) GD TD 14 C EXIT ACTION C COMPUTATION C COMPUTATION C EVAND. DPSI, LE. O., AND. ABSI, VP. EQ. ABS(VS) GD TO			700		
C			\$ 6		
12 CALL IMPACT (PHI(1) PHI(2), PSI, DPSI) C TEST FOR EXIT ACTION PHID-PHI(1)/1/22 IF (PHID-LE, 160, 0) GD TD 14 C EXIT ACTION C COMPUTATION C COMPUTATION C C COMPUTATION C C COMPUTATION C C EXIT ACTION C EXIT ACTION C EXIT ACTION C EXIT ACTION C C COMPUTATION C EXIT ACTION C	25		200		1
F (TIME GT. 5.0) GG TO 16 C TEST FOR EXIT ACTION					
C		1F (Time. Gr. 5.0) GO 7D 16	- 646		
PHID-PHI(1)/72 IF (PHID.LE.180.0) GD TD 14 C			262		ì
F (PHID-PHI(1)/22 F (PHID-LE:160.0) GO TO 14		TEST FOR	244		
HIDP-HI(1)/ZZ	245		170		
F (PHID.LE.160.0) GD TO 14		PHID=PHI(1)/22	A 246		1
C EXIT ACTION VP=0PSIs (C=COS(ALPHR)+G) IF (AHITO T=T=S) IF (AHITO T=TS) C COMPUTATION GF VELOCITIES -VP AND VS FOR ENTRANCE -ACTION IF (AHITO T=TS) C COMPUTATION GF VELOCITIES -VP AND VS FOR ENTRANCE -ACTION IF (AHITO T=TS) C COMPUTATION GF VELOCITIES -VP AND VS FOR ENTRANCE -ACTION IF (AHITO T=TS) C COMPUTATION GF VELOCITIES -VP AND VS FOR ENTRANCE -ACTION IF (AHITO T=TS) C COMPUTATION GF VELOCITIES -VP AND VS FOR ENTRANCE -ACTION C COMPUTATION GF VELOCITIES -VP AND VS FOR ENTRANCE -ACTION C COMPUTATION GF VELOCITIES -VP AND VS FOR ENTRANCE -ACTION C COMPUTATION TESTS C ENTRANCE ACTION TESTS IF (AHITO T=TS) C ENTRANCE ACTION TESTS IF (AHITO T=TS) IF (AHITO T=TS) C ENTRANCE ACTION TESTS IF (AHITO T=TS) IF (AH		IF (PHID.LE.160.0) GD TO 1	A 247		
C EXIT ACTION C COMPUTATION OF VELOCITIES VP AND VS FOR EXIT ACTION VP=0978 (10 constant and			A 248		
C COMPUTATION OF VELOCITIES VP AND VS FOR EXIT ACTION VP-DPSI (C*COS(ALPHR)+G) VS-PHI (1)-PS-COS(ALPHR)+G) VS-PHI (2)-SE-COS(ALPHR)+G) VS-PHI (1)-PS-COS(ALPHR)+G) VS-PHI (2)-SE-COS(ALPHR) VS-PHI (2)-SE-COS(ALPHR)+G) VS-PHI (2)-SE-COS(ALPHR)+G) IF (PHI(2)-GE-OAND.DPSI.LE-OAND.ABS(VP)-G7-ABS(VS)) IF (PHI(2)-GE-OAND.DPSI.LE-OAND.ABS(VP)-G7-ABS(VS)) IF (PHI(2)-GE-OAND.DPSI.LE-OAND.ABS(VP)-G7-ABS(VS)) IF (PHI(2)-LE-OAND.DPSI.LE-OAND.ABS(VP)-G7-ABS(VS)) C COMPUTATION GF-VELOGITIES-VP AND VS FOR ENTRANCE ACTION IF (PHI(2)-GE-OAND.DPSI.GE-OAND.ABS(VP)-G7-ABS(VS)) C ENTRANCE ACTION TESTS C ENTRANCE ACTION TESTS IF (PHI(2)-GE-OAND.DPSI.GE-OAND.ABS(VP)-G7-ABS(VS)) IF (PHI(2)-GE-OAND.ABS(VP)-GF-ABS(VS)) IF (PHI(2)-GE-OAND.ABS(VP)-GF-ABS(VS)) IF (PHI(2)-GE-OAND.ABS(VP)-GF-ABS(VS)) IF (PHI(2)-GE-OAND.ABS(VP)-GF-ABS(VS)) IF (PHI(2)-GE-OAND.ABS(VP)-GF-ABS		-	₽ 249		Ì
COMPUTATION OF VELOCITIES VP AND VS FOR EXIT ACTION VP=0PSI*(C*COS(ALPHR)+G) VS=PH1(2)*9**COS(PH1(1)+DSI-ALPHR) VS=PH1(2)*9**COS(PH1(1)+DSI-ALPHR) VS=PH1(2)*9**COS(PH1(1)+DSI-ALPHR) C EXIT ACTION TESTS C EXIT ACTION TESTS IF (PH1(2)*GE**O**AND**DPSI*GE**O**O**O**O**O**O**O**O**O**O**O**O**O	22		250		
VS-PHI(2) • SPCOS (ALPHR) + G) VS-PHI(2) • SPCOS (ALPHR) + G) VS-PHI(2) • SPCOS (ALPHR) FPI FPI		COMPUTATION OF VELOCITY	A 25.1		
VS-PHI(2)*B3CDS(PHI(1)-PSI-ALPHR) WRITE (6,38) VP, VS-AND-PHIIGI-LIT.4460,) GD 10 13 13 IF (ABS(ABS(VP)-ABS(VS)).LT.2.0) GD 10 1 C EXIT ACTION TESTS C EXIT ACTION TESTS C EXIT ACTION TESTS IF (PHI(2).GE.0AND.DPSI.LE.0AND.ABS(VP).LT.ABS(VS)) GD 10 1 IF (PHI(2).GE.0AND.DPSI.LE.0AND.ABS(VP).LT.ABS(VS)) GD 10 1 IF (PHI(2).GE.0AND.DPSI.GE.0AND.ABS(VP).LT.ABS(VS)) GD 10 5 IF (PHI(2).LE.0AND.DPSI.GE.0AND.ABS(VP).LT.ABS(VS)) GD 10 5 IF (PHI(2).LE.0AND.DPSI.GE.0AND.ABS(VP).LT.ABS(VS)) GD 10 5 IF (PHI(2).LE.0AND.DPSI.GE.0AND.ABS(VP).GT.ABS(VS)) GD 10 5 IF (PHI(2).LE.0AND.DPSI.GE.0AND.ABS(VP).GT.ABS(VS)) GD 10 5 C COMPUTATION OF VELOCITIES VP AND VS FOR ENTRANCE ACTION C COMPUTATION OF VELOCITIES VP AND VS FOR ENTRANCE ACTION IF (PHITOT.GT.30AND.PHITOT.LT.1450.) GD 10 15 A VS-PHI(2)*B8COS(APIN(1)*PSI.ALPHR) IF (PHITOT.GT.30AND.PPSI.GE.0AND.ABS(VP).GT.ABS(VS)) GD 10 1 A WITE (6,38) VP, VS IF (PHI(2).GE.0AND.DPSI.GE.0AND.ABS(VP).LT.ABS(VS)) GD 10 1 IF (PHI(2).LE.0AND.DPSI.GE.0AND.ABS(VP).GT.ABS(VS)) GD 70 1 IF (PHI(2).LE.0AND.DPSI.LE.0AND.ABS(VP).GT.ABS(VS)) GD 70 1 IF (PHI(2).LE.0AND.DPSI.LE.0AND.ABS(VP).GT.ABS(VS) GD 70 1 IF (PHI(2).LE.0AND.ABS(VE).LE.0AND.ABS(VP).GT.ABS(VP).GT.ABS		VP=DPSI+(C+COS(ALPHR)+G)			1
### (PMITOT.GT.30 - AND.PHITOT.CT.11166.) GD 10 13 ###################################		VS=PHI (2) + R±CDS (pHI (4) - DCI - AI DUD)	7C7 W		
##ITE (6.38) VP,VS EXIT ACTION TESTS EXIT ACTION TESTS EXIT ACTION TESTS IF (PHI(2).GE.OAND.DPSI.GE.O.) GO TO 1 IF (PHI(2).GE.OAND.DPSI.LE.OAND.ABS(VP).GT.ABS(VS)) GO TO 1 IF (PHI(2).GE.OAND.DPSI.LE.OAND.ABS(VP).LT.ABS(VS)) GO TO 1 IF (PHI(2).GE.OAND.DPSI.GT.OAND.ABS(VP).LT.ABS(VS)) GO TO 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).EQ.ABS(VS)) GO TO 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).GT.ABS(VS)) GO TO 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).GT.ABS(VS)) GO TO 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).EQ.ABS(VS)) GO TO 1 A V=DPSI.VC.OCS(ALPHR)+G) C COMPUTATION GF VELOCITIES VP AND VS FOR ENTRANCE ACTION FOR THE (6.38) VP,VS C ENTRANCE ACTION TESTS IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS)) GO TO 5 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS)) GO TO 7 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS) GO TO 7 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS) GO TO 7 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.ABS(VP).GT.AB		TE CALLY CONTINUE OF THE ABOUT THE A	A 253		
### (### #	25.5	MOTE (R. 30) VO VS	A 254		
C EXIT ACTION TESTS C EXIT ACTION TESTS C EXIT ACTION TESTS If (PHI(2) GE. 0. AND. DPSI. LE. 0. AND. ABS(VP). GT. ABS(VS)) GO TO		TE (ABC/VO)-ABC/VC)) IT 5 6) CO	4 255		
C EXIT ACTION TESTS C IF (PHI(2).GE.OAND.DPSI.GE.O.) GO 10 1 IF (PHI(2).GE.OAND.DPSI.LE.OAND.ABS(VP).GT.ABS(VS)) GO 10 5 IF (PHI(2).GE.OAND.DPSI.LE.OAND.ABS(VP).LT.ABS(VS)) GO 10 1 IF (PHI(2).LE.OAND.DPSI.LE.OAND.ABS(VP).LT.ABS(VS)) GO 10 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).LT.ABS(VS)) GO 10 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).GT.ABS(VS)) GO 10 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).EQ.ABS(VS)) GO 10 1 IF (PHI(2).LE.OAND.DPSI.GT.OAND.ABS(VP).EQ.ABS(VS)) GO 10 1 C COMPUTATION OF VELOCITIES VP.AND.VS.FOR ENTRANCE.ACTION C COMPUTATION OF VELOCITIES VP.AND.VS.FOR ENTRANCE.ACTION A VS-PHI(2).EOAND.PHITOT.LT.1450.) GO 10 15 C ENTRANCE ACTION TESTS C ENTRANCE ACTION TESTS IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).EQ.ABS(VS)) GO 10 1 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).EQ.ABS(VS) GO 10 1 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).EQ.ABS(VP).EQ.ABS(VS) GO 10 1 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).EQ.ABS(VP).EQ.ABS(VS) GO 10 1 IF (PHI(2).GE.OAND.DPSI.GE.OAND.ABS(VP).EQ		: (max(max(m) max(m)).c::(*) an	A 256		
F (PHI(2) GE.0. AND. DPSI.GE.0.) GD.10.1 F (PHI(2) GE.0. AND. DPSI.GE.0. AND. ABS(VP) GT.ABS(VS) GD.10.5 F (PHI(2) GE.0. AND. DPSI.GE.0. AND.ABS(VP).GT.ABS(VS) GD.10.1 F (PHI(2) GE.0. AND. DPSI.GT.0. AND.ABS(VP).EQ.ABS(VS) GD.10.1 F (PHI(2) GE.0. AND. DPSI.GT.0. AND.ABS(VP).GT.ABS(VS) GD.10.1 F (PHI(2) LE.0. AND.DPSI.GT.0. AND.ABS(VP).GT.ABS(VS) GD.10.1 F (PHI(2) LE.0. AND.DPSI.GT.0. AND.ABS(VP).EQ.ABS(VS) GD.10.1 F (PHI(2) LE.0. AND.DPSI.GT.0. AND.ABS(VP).EQ.ABS(VS) GD.10.1 F (PHI(2) LE.0. AND.DPSI.GT.0. AND.ABS(VP).EQ.ABS(VS) GD.10.1 F (PHI(2) LE.0. AND.PPI.GT.0. AND.ABS(VP).GT.ABS(VS) GD.10.1 F (PHI(2) GE.0. AND.DPSI.GE.0. AND.ABS(VP).GT.ABS(VS) GD.10.1 F (PHI(2) GE.0. AND.DPSI.GE.0. AND.ABS(VP).GT.ABS(VS) GD.10.1 F (PHI(2) GE.0. AND.DPSI.GE.0. AND.ABS(VP).EQ.ABS(VS) GD.10.1 F (PHI(2) GE.0. AND.DPSI.GE.0. AND.ABS(VP).EQ.ABS(VS) GD.10.1 F (PHI(2) GE.0. AND.DPSI.GE.0. AND.ABS(VP).GT.ABS(VS) GD.10.1			A 257		
F (PHI(2) GE.0AND.DPSI.LE.Q. 400.ABS(VP) GT.ABS(VS) GO TO 5			A 258		
IF (PHI(2):GE.OAND.DPSI.LE.QAND.ABS(VP).GT.ABS(VS)) GO TO TE FOR THE (PHI(2):GE.OAND.DPSI.LE.OAND.ABS(VP).LT.ABS(VS)) GO TO TE FOR THE (PHI(2):GE.OAND.DPSI.GE.OAND.ABS(VP).EQ-ABS(VS)) GO TO TE FOR THE (PHI(2):LE.OAND.DPSI.GT.OAND.ABS(VP).EQ-ABS(VS)) GO TO TE FOR THE (PHI(2):LE.OAND.DPSI.GE.OAND.ABS(VP).EQ-ABS(VS)) GO TO TE FOR THE (PHI(2):GE.OAND.PHITOT.LT.1450.) GO TO 15 C COMPUTATION OF VELOCITIES VP AND VS FOR ENTRANCE ACTION A VS=PHI(2):BS-COS(PHI(1)-PSI-ALPHR) IF (PHI(2):BS-COS(PHI(1)-PSI-ALPHR) C ENTRANCE ACTION TESTS C ENTRANCE ACTION TESTS C ENTRANCE ACTION TESTS C FORTICO:GE.OAND.DPSI.GE.OAND.ABS(VP).EQ-ABS(VS)) GO TO TE FOR TE FOR TE FOR THE (PHI(2):GE.OAND.DPSI.GE.OAND.ABS(VP).EQ-ABS(VS)) GO TO TE FOR TE FOR THE (PHI(2):GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS)) GO TO TE FOR TE FOR TE FOR THE COAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS) GO TO TE FOR TE FOR TEXTS IF (PHI(2):GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS) GO TO TE FOR TEXTS IF (PHI(2):GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS) GO TO TE FOR TEXTS IF (PHI(2):GE.OAND.DPSI.GE.OAND.ABS(VP).GT.ABS(VS) GO TO TE FOR TEXTS IF (PHI(2):GE.OAND.DPSI.LE.OAND.ABS(VP).GT.ABS(VS) GO TO TEXTS IF (PHI(2):GE.OAND.DPSI.LE.OAND.ABS(VP).GT.A	260	1F (PHI12).GE.0AND. DBS I. GE.0. CD	A 259		
F (PHI(2):GE:0AND.OPSI:LE:0AND.ABS(VP):Eq.ABS(VS))) 	(OHI (2) CE D AND DEET IE O AND ADDITION OF THE OHIO	98.4		1
F (PHI(2):LE:0AND.DPSI:LE:0AND.ABS(VS)) GO TO 1		(MILLS) (SELECTION OF THE SELECTION OF T	A 261		
F (PHI(2) G G G G G G G G G		(Fritch of the Control of the Contro	A 262		
IF (PHI(2):LE: 0AND. DPSI:GT: 0AND. ABS(VP):LT.ABS(VS)) GD TO 5 IF (PHI(2):LE: 0AND. DPSI:GT: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 IF (PHI(2):LE: 0AND. DPSI:GT: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 C COMPUTATION OF VELOCITIES VP AND VS FOR ENTRANCE ACTION A VS=PHI(2):B*COS(ALPHR)+G) IF (PHITOT.GT:30AND.PHITOT.LT:1450.) GD TO 15 WRITE (6,38) VP.VS IF (PHI(2):GE: 0AND. PPSI:GE: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:GE: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:GE: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:GE: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):EQ.ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. DPSI:LE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1 IF (PHI(2):GE: 0AND. ABS(VP):GT: ABS(VS)) GD TO 1		(THILE): GE: G:: AND I DEST: EE: G: AND: AND (VP): EQ: AND (VS) } - GO	A 263		1
IF (PHI(2):LE.0AND.DPSI.GT.0AND.ABS(VP).GT.ABS(VS)) GO TO 1 IF (PHI(2):LE.0AND.DPSI.GT.0AND.ABS(VP).GT.ABS(VS)) GO TO 1 IF (PHI(2):LE.0AND.DPSI.LE.0.) GO TO 5 C COMPUTATION OF VELOCITIES VP AND VS FOR ENTRANCE ACTION A VS=PHI(2)*B*COS(ALPHR)+G) VS=PHI(2)*B*COS(ALPHR)+G) VS=PHI(2)*B*COS(ALPHR)+G) VS=PHI(2)*B*COS(ALPHR)+G) ANTITE (6,38) VP,VS C ENTRANCE ACTION TESTS A F (PHI(2):GE.0AND.DPSI.GE.0AND.ABS(VP).GT.ABS(VS)) GO TO 5 IF (PHI(2):GE.0AND.DPSI.GE.0AND.ABS(VP).EQ.ABS(VS)) GO TO 1 IF (PHI(2):GE.0AND.DPSI.GE.0AND.ABS(VP).EQ.ABS(VS)) GO TO 1 IF (PHI(2):GE.0AND.DPSI.GE.0.) GO TO 1 IF (PHI(2):GE.0AND.DPSI.LE.0.) GO TO 1	1	(MI(2):LE:0:.AND:DPSI:GT:0:.AND.ABS(VP):LT.ABS(VS)) GO	A 264		
IF (PHI(2):LE:0:.AND.DPSI:GT:0.:AND.ABS(VP).EQ.ABS(VS)) GO TO 10 C COMPUTATION OF VELOCITIES VP AND VS FOR ENTRANCE ACTION A VP=DPSI*(C+COS(ALPHR)+G) IF (PHITOT.GT:30AND.PHITOT.LT:1450.) GO TO 15 A MRITE (6,38) VP,VS C ENTRANCE ACTION TESTS C ENTRANCE ACTION TESTS IF (PHI(2):GE:0AND.DPSI:GE:0AND.ABS(VP).GT:ABS(VS)) GO TO 1 IF (PHI(2):GE:0AND.DPSI:GE:0AND.ABS(VP).EQ.ABS(VS)) GO TO 1 IF (PHI(2):LE:0AND.DPSI:LE:0AND.ABS(VP).EQ.ABS(VS)) GO TO 1 IF (PHI(2):LE:0AND.ADSI:LE:0AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ADS.AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ADD.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ADD.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ABS(VB).AND.ADD.ABS(VB).AND.AB	B	(PH1(2).LE.OAND.DPSI.GT.OAND.ABS(VP).GT.ABS(VS)) GO	A 265		
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	IF (PHI(2).LE.OAND.DPSI.LE.OAND.ABS(VP).EQ.ABS(VS)) GO TO 1	A 286			
	WRITE (6.39) FORMAX EFFORMAX FERNINA	A 287			
	STOP STOP	A 288			
290		A 289			
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	FUNDAL (7F10.5)	200			
208	TUKMA! (1H1,5X,2HA=,F13	A 294			
	1EX= , F9.4	200			
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	FORMAT	A 307			
	28 FORMAT (2F10.2)	# 308 #			
	29 FORMAT (1H ,5X,8HPSUBD1 =, F5.1,3X,8HPSUBD2 =, F5.1,7/6X,5HMS1 = F4.0	808			
		A 310			
	2.5.3X,8HCAPRP2 =,F8.5//6X,5HRP2 =,F8.5.3X,5HBP3 = F8 E//CY PILET	115 4			
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330	END	A 329			
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1662	LALL INS (PHI(2), PHII, DELPHI, VSI, PSI, DPSI, AGNE, BGNE, CONE, DONE, AA1	ם ט נ		
- AAA-	144. A44. AA5. AA6. AA7. AA8. AA9. AA10. AA11. AA12. AA14. AA15. AA18.	*		
24420	24.25. 1.4.4. G.AA.19. AA.20. AA.21. AA.22. AA.23. AA.24. AA.25. AA.26. AA.27. AA.29. AA.29.	·		
4443. AAA3	**************************************	50		
3 41	00010 65 0 1 10			
	10	C 22		
		ı		
	(PHI(2)*DPHI2.LT.0.) T1D=T1=WHOT0(AA26+AA30)			
) <u>1</u>	_1 1R=0			
) II	(IPR.LT.0.) IPR=0.			
AA45	AA45=13*AA18+1PR*AA17*U-(12*N32-AA42*11R*N31/AA31)*AA15*AA18/AAAA	C 27		
144)	AA46erPR+AA17+AA17+AA21+UeU+AA15+AA18+AA34+AA42+N31+R31/{AA3+AA	ه ۹ پ د		
AA47.	AAG7=9 +DM9+11/7+4190+11/7+41/7+41/7+41/7+41/7+41/7+41/7+41/7			
1444		C 31		
AA48:	4	1		- [
1AA18	1AA18+13-0M2+AA17+AA19			
AA49	AA49=AA15+AA18+AA35+AA42/{AA31+AA44}	က 34		
AA50:	AA50=MP+RCP+R4+AA17+DM2	l		
IHdo	DPHI(1)=PHI(2)	96 2		
1440	DPHI(2)=("AA46+PHI(2)+PHI(2)+AA47+PHI(2)+AA48-AA49+SIN(PHI1C+HA44+	ج د د		
	DETILO: *22+PHI(1)-PHIPR*ZZ))+AA50*SIN(GAMAPP-PSI-PSIC))/AA45			ı
CKS		0 6		
				- [

SUBROUTINE OUTP (T, PHI, DPHI, THLE SUBROUTINE OUTP (T, PHI, DPHI, 12). DIMENSION PHI(2), DPHI(2), DPHI(2), DPHI(2), LITA, PHI(10, PHIRR, N31, M22, DMEGA, DBAS, M33, CAPHA2, N13, CAPHA2, R32, DMEA, B3, HAP, N13, CAPHA2, R31, M2, M11, M2, DMEA, DPHI ZPZ, N13, CAPHA2, N1, 12, CAMMA2, BB3, HB2 ALTA3, DELLA4, BETA1, BETA2, BETA1, BETA2, BETA1, DPHICA, DPHI COLEMNY, FIZMAX, FIZMAX, FIZMAX, DNM COLEMNY, PHIDD-PHIPR PHIDPHI(1)/ZZ DELPHI PHIDD-PHIPR PHIDPHI(1)/ZZ DHIPPHIDT-PHIDT-CELPH, CG, PI, V, VSI, 100 CALL KINEM (A, B, ALDHR, PHI, C, G, P, V, V, VSI, MA2, AA3, AA4, AA5, AA6, AA7, AA8, AA9, AA5, AA6, AA7, AA8, AA9, AA5, AA4, AA5, AA6, AA7, AA8, AA9, AA7, AA8, AA4, AA5, AA6, AA7, AA8, AA7, AA8, AA4, AA5, AA6, AA7, AA8, AA7, AA8, AA4, AA7, AA7, AA8, AA8, AA8, AA7, AA8, AA8				
DIRECTOR WILL STATE AND ST	-	ation a sur thou the th etino		
CDMRCH S.G. W. A.P.H. P. I. Z., M. I. W. B., B. I. I. I. I. I. I. P. EEST L. LANDON, C. C. C. A.P.H. P. I. Z., M. I. W. B., B. I. I. I. I. I. P. EEST L. LANDON, C.		M3.MP.11.12.		
COMPON A, B, C, R, ALHR RP, IZ, IZ, III, MZ, MZ, B, II, IZ, IZ, IP, EEST, MGI, MGZ, 1111. PHILD: PHILD: MX1. MZ, CMEGA.ODZ, BCIT, PHILC, IEST, MGI, MGZ, 3043, MDP, CAPEB, C, CREBZ, RES, MRS, S, CHRAN, T, IFT, Z, R, R, R, S, S, G, R,		DPHI		
22 NO. 14 10. PHILOL. PRINCE, CARLS, CARNAL, CARLS, CARNAL, CARLS, CARLS	•			
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4.143, 10.1442, 10.17, 173, 10.10, 10.10, 11.10, 4.11.10, 4.11.210, 4.12.		ara, wrs. Lafkbi, Lafkbi, kbi, kbi, iheiai, theiai, ki, ki, ki, ka, ka, khoi, khoi, khoz, r 3ho3, rhop, Ji, Ji, Gammaz, Gamasp, Gammas, Gammaa, Gammaa, Gammabp, Dei Tas, se		
SA. A. MANAL, IN. 172, 172, MI. MUI. RCP. ISSIC.512, 54, 55, A1, A2, DPHIZ, DPSIZ, F G G MAX, FIZAMA, FIZAMA, PRINKA, PI. ALDHEW, AL		41743. DELIA4. BETA1. BETA2. BETA3. D1. D2. A111N. A11FIN. A12FIN. A12		
COLONIA, 17 12043, 17 12043, 17 12043, PURIOLID DEFICION ACTALLY 18, 12043, 17 12043, PURIOLID DEFICION ACTALLY 19 10 10 10 10 10 10 10 10 10 10 10 10 10	•	5A1. ALPHA2. IN. T2. T3. MJ. MUI, RCP, PSIC, S1, S2, S4, S5, A1, A2, DPHI2, DPSI2, F		
PHID PHI	2	TOWAX, PNMAX, PN, ALPHE	***	
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PHITOT-PHITOT-DELPH PHITOT-PHITOT-DELPH PHITOT-PHITOT-DELPH PHITOT-PHITOT-DELPH PHITOT-PHITOT-DELPH Value		OHI PER PHYSICAL CONTROL OF THE PHYSICAL CONTROL OF TH	•	
PHILIPPHILOT: 22 (ALL KINEM (A.B.ALPHR.PHI.C.G.P.Q.S.PSI.DPSI.ADNE_BONE_CONE_CONE_LOWE_U (A.V.ST) (ALL AND APALABA ANG AATA AND AAZI.ASLABSA AATO AATI.ANIL AATI.ANIS_AATI (AAZ AA3.AA4.ABA.ABC.AAZI.AAZI.AAZA.AAZI.AAZI.AAZI.AAZI.AAZ	15	PHI TOT = PHI TOT + DEL PHI	† '	
CALL KINEM (A. B. ALPHR, PHI, C.G. P. Q. S. PSI, DPSI, ADDE, BONE, CONE, DATE, V. VST) CALL LING (PHI(2), PHIT, DELPHI, VST, PSE, PDSI, ADDE, ADDE, CONE, DANE, U. V. VST) CALL LING (A. A. A		PHIT=PHITO1*22	- ,	
CALL ING (PHI(2), PHIT, DELPHI, UST, PST, DPSI, ADDSI, ADNE, BONE, CONE, UD CALL ING (PHI(2), PHIT, DELPHI, UST, PST, DPSI, ADDSI, ADNE, BONE, CONE, UD CALL ING (ANA), AAG, AAG, AAG, AAG, AAG, AAG, AAG, A		38±0	- •	
1, V, V31) CALL IN2 (PHI(2), PHIT DELPHI, VST, PST, PDST, ADME, BONE, COME, DONE, AA1, BA2, AA3, AA4, AA5, AA4, AA5, AA2, AA2, AA2, AA3, AA4, AA5, AA2, AA2, AA2, AA2, AA3, AA4, AA5, AA2, AA3, AA4, AA5, AA2, AA3, AA4, AA5, AA3, AA4, AA5, AA3, AA4, AA4		NEM (A.B,ALPHR,PHI	† •	
AA2 AA3 AA4, AA5, AA6, AA7, AA8 AA9, AA7, AA8, AA9, AA1, AA12, AA13, AA14, AA5, AA18 AA2 AA3, AA4, AA5, AA6, AA7, AA2, AA2, AA2, AA26, AA27, AA86, AA27, AA89, AA49, AA39, AA44, AA5, AA64, AA5, AA64, AA5, AA64, AA5, AA64, AA5, AA64, AA5, AA64, AA66, AA66, AA64, AA64, AA66,			- ,	
1 MA2, AA3, AA4, AB5, AA6, AA7, AA8, AA9, AA1, AA12, AA11, AA12, AA13, AA14, AA8, AA4, AA17, AA18, AA31, AA32, AA21, AA23, AA24, AA22, AA24, AA38, AA39, AA44, AA4	88	CALL INS (PHICS), PHIT DELPHI VSI DSI ADNE BONE COME AND	- (
2. AA77. AA18. AA19. AA20. AA21. AA22. AA23. AA24. AA22. AA26. AA27. AA39. AA3		1AA2.AA3.AA6.AA5.AA6.AA7.AA8.AA9.AA41. AA41. AA42. AA44. AA44. AA44.	i	
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4443, A444 IF (DPSI-DPSI2.GE.O.) IPR=IP+AA22 IF (DPSI-DPSI2.LT.O.) IPR=IP+AA22 IF (DPSI-DPSI2.LT.O.) IRR=II+ABS(MU)*RHOI*(AA26+AA30) IF (PHI(2)*DPHI2.GE.O.) IRR=II+ABS(MU)*RHOI*(AA26+AA30) IF (PHI(2)*DPHI2.LT.O.) IRR=II+ABS(MU)*RHOI*(AA26+AA30) AA45=IR*A17*A417*A419 AA46=IRR*A17*A417*A419 AA50=MR*RCP*RCP*A4A17*A419 AA50=MR*RCP*RCP*RCP*RA4A17*A42 AA50=MR*RCP*RCP*RCP*RCP*RCP*RCP*RCP*RCP*RCP*RC		38830 -8831 - 8832 - 8833 -8835 - 8835 - 8832 - 883		
If (DPSI *DPSI2.CE.O.) IPR=IP+AA22		4A40. AA40		
F (DPSI + DPSI = 1 + 0 + 0	x	DPS12 GF.0.) 1		
F (PHI(2)*DPHI2.GE.0.) IIR=I1-ABS(MU)*RHO1*(AA26+AA30) F (HYL(2)*DPHI2.LT.0.) IIR=I1-ABS(MU)*RHO1*(AA26+AA30) F (HYL(2)*DPHI2.LT.0.) IIR=I1-ABS(MU)*RHO1*(AA26+AA30) AA45=13*AA18+IPR*AA17*U-(12*N32-AA42*I1R*N31)*AA15*AA18*AA39*AA44*DA46=IPR*AA17*V+AA17*AA20*U/ABS(UMEGA)+AA15*AA18*AA39*AA42*N31/(AA31*AAD-144) AA46=PR*AA17*AA17*AA17*AA17*AA19 AA48=-AA15*AA17*AA17*AA19*AA19 AA48=-AA15*AA17*AA17*AA19 AA48=-AA15*AA18*AA17*OM2 AA49=-AA15*AA17*AA17*OM2 DPHI2.*(-AA43*AA17*AA19*AA17*OM2 DPSI2-U*DPHI2*Y*PHI(2)*PHI(2)*AA7*PHI(2)*AA89*SIN(PHI1C*N31*(PM DDPI2*U*PHI(1)*PHI(2)*PHI(2)*AA3*AA2*N31*PHIT)/AA31*(PM DDPI2*U*PHI(2)*PHI(2)*AA3*AA2*N31*PHIT)/AA31*(BM DDPI2*U*PHI(2)*PHI(2)*AA31*AA33*AA2*N31*PHIT)/AA31*(BM DDPI2*U*PHI(2)*PHI(2)*AA31*AA33*AA2*N31*PHIT)/AA31*(BM DDPI2*U*PHI(2)*PHI(2)*AA31*AA35*AA42*N31*PHIT)/AA31*(BM DDPI2*U*PHI(2)*PHI(2)*AA31*AA35*AA42*N31*PHIT)/AA31*(BM DDPI2*U*PHI2*U*AA116*AA33*AA2*N31*PHIT)/AA31*(BM DDPI2*U*DPI2*U*AA116*AA35*AA42*N31*PHIT)/AA31*(BM DDPI2*U*DPI2*U*AA116*AA35*AA42*N31*PHIT)/AA31*(BM DDPI2*U*DPI2*U*AA116*AA35*AA42*N31*PHIT)/AA31*(BM DDPI2*U*D		(DPSI+DPSI2.LI.O.)		
F (PHI(2)*DPHI2.LT.0.) IP=11-ABS(MU)*RND1*(AA26+AA30) DFR-0.		(PHI(2)*DPHI2.GE.0.)	i	
		(PHI(2)*DPHI2.LT.0.)		
The control of contro		(118.LT.0.) -118.0.	- 1	
A45=13*AA B+IPR*AA17*U-(12*N32-A42*11R*N31)*AA31)*AA15*AA18/A44 144) A46=IPR*AA17*V+AA17*AA21*U-U+A415*A518*AA42*A31*A31*AA A47=2.*CM2*AA17*AA21*U-U+A415*A518*AA42*A33*AA2*N31/(AA31*A D 144) A449=-A415*A418*A32*A442/(AA31*A44)-A415*A418*AA43*T2/AA44-A416*D 1AA18*T3-CM2*AA17*A419 AA50-MP*RCP*R4*AA17*DM2 DPHI2=(-A46*PH1(2)*PH1(2)-A47*PH1(2)+AA48-A449*SIN(PH1IC+N31*(PH D DPSI2=U+DPH12+V*PH1(2)*PH1(2) C COMPUTATION OF CONTACT FORCES- C COMPUTATION OF CONTACT FORCES- 131*N31*PH1(2)*PH1(2)/AA31-T2*AA42-N31*PH1(2)/AA31-AA42*AA34*N 131*N31*PH1(2)*PH1(2)/AA31-T2*AA43-I2*N32*DPH12)/AA42 F12=(F23*G1*F23*AA15-T3*AA15)/AA17 F12=(F23*G1*F23*AA15-T3*AA15)/AA17 F13=(F23*G1*F23*AA15-T3*AA15)/AA17 F14=(F23*G1*F23*AA15-T3*AA15)/AA17 F17=(F13*DPH12+F23*AA15-T3*AA15)/AA17 F18-F19-MSA1*NANANANANANANANANANANANANANANANANANAN	90	.0.) IPR=0.		
AA47=2.+CM2*AA17*AA20*U/AB5(CMEGA)+AA15*AA18*AA34*AA2*A31/(AA31*AB)		18+IPR+AA17+1		
A447-2.*CM2*A417*AA20*U/ABS(CMEGA)+AA15*AA18*AA42*N31/(AA31*A D 1444) A448-A415*AA18*AA32*AA42/(AA31*AA44)-AA15*AA18*AA33*AA42*N31/(AA31*A D 1AA18*T3-CM2*AA17*AA19 AA50=MP*RCP*R4*AA17*CM2 DPH12=(-AA46*FHI(2)*PHI(2)-AA47*PHI(2)+AA48-AA49*SIM(PHI1C*N31*(PH D 1100*Z2*PH)(1)-PHIPR*Z2)+AA50*SIM(GAMAPP-PSI-PSIC))/AA49- C CCMPUTATION OF CONTACT FORCE9 E 73=(-AA32*AA42/AA31-T2*AA43-AA33*AA42*N31*PHI(2)/AA31-AA42*AA31*DHI2)/AA44 Z32-AA42*IIR*N31/AA31)+DPHI2)/AA44 E 73=(-AA32*AA44+T2*AA43-12*N32*DPHI2)/AA42 IF (F23.GI-F23*AA15-T3*AA15)/AA77 F 12=(F23.GI-F23*AA15-T3*AA15)/AA77 F 12=(F23.GI-F23*AA15-T3*AA15)/AA77 F 142-GI-F12MAX) F 12MAX=F 12 IF (F12.GI-F12MAX) F 12MAX=F 12 IF (F12.GI-F12MAX) F 12MAX=F 12 IF (FN.GI-PMBAX) P NUMAX=F 12 IF (FN.GI-PMBAX) P NUMAX=PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		0=!PK+AA+7+V+AA+7+A	1	
A448		7-7 +0402-6403-6403-		
A448=-A415*AA18*AA32*AA42/(AA31*AA44)-AA15*AA18*AA5*T2/AA44-AA18*D A449=A415*AA17*AA19 A450=MP*RCP*R4*AA17*DM2 DPA12=(PA46*PHI(2)*PHI(2)-AA47*PHI(2)+AA9*SIN(PHIIC+N31*(PHDDPIIZ)-PHIIC2)*PHI(2)+PHI(2)*PHI(2)+PHI(2)*PHI(2)+PHI(2)*PHI(2)+PHI(2)*AA42*AA42*AA34*D C COMPUTATION OF CONTACT FORCES C COMPUTATION OF CONTACT FORCES F 23=(-AA32*AA42/AA31)*DPHI2)/AA44 F 12=(F 23*AA42*IR*N31/AA31)*DPHI2)/AA44 F 12=(F 23*AA42*IR*N31/AA31)*DPHI2)/AA44 F 12=(F 23*AA44*IZ*AA43*IZ*N3*DPHI2)/AA42 F (F 12:GT.F 12MAX) F 12MAX = F 12 IF (F 12:GT.F 12MAX) F 12MAX = F 12 IF (PN:GT.PNMAX) PNMAX = F 12 IF (PN:GT.PNMAX) P	35	-C CMC - AA / - AA CU - U/ AB 3 (CME CA) + AA 15 + AA 18 + AA 33 + AA 2 + N3 1 / (AA 31		
AA49=AA15-AA17-AA19) }	=-44/5+44/8+48/30+48/4//44/1+44/4/	1	
AA59=AA15+AA18+AA35+AA42/(AA31+AA49) AA50=MP+RCP+R4+AA17+0M2 DPHIZ=(-AA46+PHI(2)+PHI(2)-AA7+PHI(2)+AA69+SIN(PHIIC+N31*(PH) DPHIZ=(-AA46+PHI(2)+PHI(2)-PHI(2)-AA7+PHI(2)+AA49+SIN(PHIIC+N31*(PH) DPSIZ=U*DPHIZ+Y*PHI(2)+PHI(2) C COMPUTATION OF CONTACT FORCES C COMPUTATION OF CONTACT FORCES D 131*N31+PHI(2)+PHI(2)+AA35+AA42+N31+PHI(2)/AA31-AA42*AA31-AA42*AA44 E 73=(-AA32*AA42/AA31-AA35*AA42*SIN(PHIIC+N31*PHII)/AA31+(12*N) D 131*N31+PHI(2)*PHI(2)/AA31+DPIIZ)/AA44 F 12=(F23*AA44+12*AA43-12*AA15-13*AA16)/AA17 D N=(-I3*DPHIZ+F23*AA15-13*AA16)/AA17 IF (F23.GT.F23*AA415-T3*AA16)/AA17 D N=(-I3*DPHIZ+F23*AA15-T3*AA16)/AA17 D N=(-I3*DPHIZ+AA31+DPSI*DPSI+2*AA20/ABS(OMEGA)+OM2*AD D N=(-I3*DPHIZ+AA21+DPSI*DPSI+2*AA20/ABS(OMEGA)+OM2*AD D N=(-I3*DPHIZ+AA21+DPSI*DPSI+2*AA20/ABS(OMEGA)+OM2*AD C TEST FOR CONTINUATION OF COUPLED MOTION		mis/ (nas - naid - na D* na 8 * na 4 5 * Z/ na 4 4 - na		
AA50=MP*RCP*R4*AA17*CM2 DPHI2=(-AA46*PHI(2)*PHI(2)-AA47*PHI(2)+AA49*SIN(PHIIC+N31*(PH D D D)) 11TOT*ZZ*PHI(1)-PHIPR*ZZ)}*AA50*SIN(GAMAPP-PSI-PSIC)}*AA49 DPSI2=U*DPHI2+V*PHI(2)*PHI(2)*DHI(2) C COMPUTATION OF CONTACT FORCES 131*N31*PHI(2)*PHI(2)*A31-T2*AA43-AA32*AA42*N31*PHI(2)/AA31-AA42*AA34*N D D T31*N31*PHI(2)*PHI(2)/AA31-DPHI2)/AA44 F12=(AA32*AA44+T2*AA43-T2*AA16)/AA44 F12=(F23*AA44+T2*AA43-T2*AA16)/AA17 IF (F12.GT.F12MAX) F12MAX=FN PN=(-I3*DPHI2+F23*AA15-T3*AA16)/AA17 IF (FN.GT.PNMAX) PNMAX=FN PNPSIC (IPR*OPSI2+AA21*DPSI*DPSI+2*OM2*DPSI*AA20/ABS(OMEGA)+OM2*A D D D D D D D D D D D D D D D D D D D		3	9	
DPHI2=(-AA46*PHI(2)*PHI(2)+AA7*PHI(2)+AA48-AA49*SIN(PHI1C+N31*(PH 11TOT*ZZ*PHI(1)-PHIPR*ZZ))+AA50*SIN(GAMAPP-PSI-PSIC))/AA45 DPSIZ=U*DPHI2+V*PHI(2)*PHI(2) C COMPUTATION OF CONTACT FORCES E F23=(-AA32*AA42/AA31-T2*AA43-AA42*N31*PHI(2)/AA31-AA42*AA34*N 131*N31*PHI(2)*PHI(2)/AA31-AA35*AA42*N31*PHI(2)/AA31+{IZ*N} D F12=(F23*AA44+T2*AA43-I2*N32*DPHI2)/AA44 F12=(F23*GI-F23*AA4+T2*AA43-I2*N32*DPHI2)/AA42 F12=(F23*GI-F23*AA4+T2*AA43-I2*N32*DPHI2)/AA42 F12=(F23*GI-F23*AA4+T2*AA43-I2*N32*DPHI2)/AA42 F12=(F23*GI-F23*AAX) F12MAX=F2 IF (F12.GI-F12MAX) F12MAX=F2 IF (F12.GI-F12MAX) F12MAX=F2 IF (F12.GI-F12MAX) F12MAX=F3 IF (F13.GI-PNAX) PNMAX=PSI*DPSI*DPSI*DPSI*DPSI*DPSI*DPSI*DPSI*D			1	
11TOT*ZZ*PHI(1)-PHIPR*ZZ)	\$	(2)-AAA7+DH1(3)+AAA6-AAA0+S1W(BU1464-44		
DPSIZ=U+DPHIZ+V+PHI(2) +PHI(2) C COMPUTATION OF CONTACT FORCES E 723=(-AA32+AA42/AA31-T2+AA43-AA42+N31+PHI(2)/AA31-AA42+AA34+N T31*N31*PHI(2)*PHI(2)/AA31-AA35+AA42*SIN(PHITC+N31*PHIT)/AA31+(12*N T32-AA42*IIR+N31/AA31)*DPHIZ)/AA42 F12=(F23+AA42+12+AA43-12+N32*DPHIZ)/AA42 F12=(F23+AA42+12+AA43-12+N32*DPHIZ)/AA42 PN=(-I3*DPHIZ+F23*AA15-T3*AA16)/AA17 IF (F23.GT.F23*AA) F12*NAX=F12 IF (F12.GT.F12*NAX) F12*NAX=F12 IF (F12.GT.F12*NAX) F12*NAX=F12 IF (FN2*GT.F12*NAX) PNMAX=PN PNPSI=(IPR*DPSIZ+AA21*DPSI*DPSI+2.*DM2*DPSI*AA20/ABS(OMEGA)+OM2*A C TEST FOR CONTINUATION OF COUPLED MOTION		3 +48450+6 1846-89-96-1-0617-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-		
C COMPUTATION OF CONTACT FORCES— E 23=(-AA32*AA42/AA31-T2*AA43-AA33*AA42*N31*PHI(2)/AA31-AA42*AA34*N D 131*N31*PHI(2)*PHI(2)/AA31-AA35*AA42*SIN(PHITC*N31*PHIT)/AA31+(12*N D 232-AA42*I1R*N31/AA31)*DPHI2)/AA42 E 12=(F23*AA44+12*AA43-12*N32*DPHI2)/AA42 PN=(-I3*DPHI2+F23*AA15-T3*AA16)/AA17 F (F23*GT.F23*AA) F 12*MAX=F12 IF (F23*GT.F23*MAX) F 12*MAX=F12 IF (F23*GT.F23*MAX) PNMAX=F12 IF (F12*GT.F12*MAX) PNMAX=PN PNFSI=(IPR*DPSI2+AA21*DPSI*DPSI*2*OM2*DPSI*AA20/ABS(OMEGA)+OM2*AA D 119-MP*RCP*R4*OM2*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		DPSI2=U*DPHI2+V*PHI(2)*PHI(2)		
C COMPUTATION OF CONTACT FORCES— E 73=(-AA32*AA42/AA31-72*AA43-AA33*AA42*N31*PHI(2)/AA31-AA42*AA34*N D 131*N31*PHI(2)*PHI(2)/AA31-AA35*AA42*SIN(PHITC*N31*PHIT)/AA31+(12*N D 232-AA42*I1R*N31/AA31)*DPHI2)/AA44 F 12=(F23*AA44+12*AA43-12*N32*DPHI2)/AA42 PN=(-I3*DPHI2+F23*AA15-T3*AA16)/AA17 D F (F23.GT.F23MAX) F 12MAX=F12 IF (F23.GT.F23MAX) F 12MAX=F12 IF (F12.GT.F12MAX) F 12MAX=F12 IF (FN-CONTINUATION OF COUPLED MOTION C TEST FOR CONTINUATION OF COUPLED MOTION				
C F23=(-AA32*AA42/AA31-T2*AA43-AA33*AA42*N31*PHI(2)/AA31-AA42*AA34*N D 131*N31*PHI(2)*PHI(2)/AA31-T2*AA42*AA34*N D 131*N31*PHI(2)*PHI(2)/AA31-AA42*AA34*N D 232-AA42*IIR*N31/AA31)*DPHI2)/AA44 E12=(F23*AA44+T2*AA43-I2*N32*DPHI2)/AA42 PN=(-I3*DPHI2+F23*AA15-T3*AA16)/AA17 IF (F23*GT.F23*AA15-T3*AA16)/AA17 IF (F23*GT.F12MAX) F12MAX=F13 IF (F12.GT.F12MAX) F12MAX=F12 PNPSI=(IPR*DPSI2+AA21*DPSI*DPSI+2.*OM2*DPSI*AA20/ABS(OMEGA)+GM2*AA D 119-MP*RCP*R4*OM2*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		COMPUTATION OF CONTACT		į
F23=(-AA32*AA42/AA31-T2*AA43-AA33*AA42*N31*PHI(2)/AA31-AA42*AA34*N D	45			
131*N31*PHI(2)*PHI(2)/AA31-AA35*AA42*SIN(PHITC+N31*PHIT)/AA31+(I2*N**) 232-AA42*IIR*N31/AA31+DHI2)/AA44 F12=(F23*AA44+T2*AA43-12*N32*DPHI2)/AA42 PN=(-I3*DPHI2+F23*AA15-T3*AA16)/AA17 IF (F23*GT.F23*AA15-T3*AA16)/AA17 IF (F23*GT.F23*AA15-F23*AA15-F23 IF (F12.GT.F12MAX) F12MAX=F12 IF (F12.GT.F12MAX) F12MAX=F12 PNPSI=(IPR*DPSIZ+AA21*DPSI+2.*OM2*DPSI*AA20/ABS(OMEGA)+OM2*AD 119-MP*RCP*R4*OM2*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		F23=(-AA32+AA42/AA31-T2+AA43-AA33+AAA2+N31+bHI(2)/AA31-AAA9+AA3		
232-AA42*11R*N31/AA31)*DPH12)/AA44 F12=(F23*AA44+T2*AA43-12*N32*DPH12)/AA42 PN=(-13*DPH12+F23*AA15-T3*AA16)/AA17 IF (F23.GT.F23MAX) F23MAX=F23 IF (F12.GT.F12MAX) F12MAX=F12 IF (FN.GT.PNMAX) PNMAX=F12 PNPSI=(IPR*DPSI2+AA21*DPSI+2.*OM2*DPSI+AA20/ABS(OMEGA)+OM2*AD C TEST FOR CONTINUATION OF COUPLED MOTION		131*N31*PHI(2)*PHI(2)/AA31-AA35*AA42*SIN(PHII[4NA1*PHII]*AA31*II		
F12=(F23*AA44+T2*AA43-12+N32*DPH12)/AA42 PN=(-I3*DPH12+F23*AA15-T3*AA15)/AA17 IF (F23.GT.F23MAX) F2MAX=F23 IF (F12.GT.F12MAX) F12MAX=F12 IF (F12.GT.PNMAX) PNMAX=F12 PNFSI=(IPR*DPSI2+AA21*DPSI*DPSI+2.*OM2*DPSI*AA20/ABS(OMEGA)+OM2*AA D 119-MP*RCP*R4*OM2*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		232-AA42+11R+N31/AA31)+DPH12)/AA44		
PN=(-I3+DPHI2+F23+AA15-T3+AA16)/AA17 IF (F23.GT.F23MAX) F23MAX=F23 IF (F12.GT.F12MAX) F12MAX=F12 IF (F12.GT.F12MAX) F12MAX=F12 IF (FN2.GT.F12MAX) F12MAX=F12 PNFSI=(IPR+OPSI2+AZ1+DPSI+2.+OM2+OPSI+AA20/ABS(OMEGA)+OM2+AA D D D D D D D D D D D D D D D D D D		F12=(F23+AA44+T2*AA43-I2*N32*DPH12)/AA42		
IF (F23.GT.F23MAX) F23MAX=F23 IF (F12.GT.F12MAX) F12MAX=F12 IF (PN.GT.F12MAX) PNMAX=F12 IF (PN.GT.F12MAX) PNMAX=PN PNPSI=(IPR*DPSI2+AA21*DPSI*DPSI+2.*CM2*DPSI*AA20/ABS(OMEGA)+CM2*AA D 119-MP*RCP*R4*CM2*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION	25	PN=(-13+DPH12+F23+AA15-T3+AA15)/ALT		
IF (F12.GT.F12MAX) F12MAX=F12 IF (PN.GT.FNMAX) PNMAX=PN FNPSI=(IPR*DPSI2+AA21*DPSI*DPSI+2.*OM2*DPSI*AA20/ABS(OMEGA)+OM2*AA D 119-MP*RCP*R4*OM2*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		IF (F23.GT.F234AX) F234AX=E23		
TF (PN.GT.PNMAX) PNMAX=PN PNPSI=(IPR*DPSI2+AA21*DPSI+2PSI+2.*CM2*DPSI*AA20/ABS(CMEGA)+CM2*AA D 119-MP*RCP*R4*CM2*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		(F12.GT.F12MAX) F12M		
PNPSI=(IPR*DPSIZ+AAZ1*DPSI*DPSI+2.*OMZ*DPSI*AAZ0/ABS(OMEGA)+OMZ*AA D 119-MP*RCP*R4*OMZ*SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		IT (PA.GT. DARAX DARAX NA	- 1	
119-MP+RCP+R4-0M2+SIN(GAMAPP-PSI-PSIC))/AA18 C TEST FOR CONTINUATION OF COUPLED MOTION		PNPSI = (IPR*DPSI24AA		
TEST FOR CONTINUATION OF COUPLED MOTION	12	-		
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	3HO3, RHOP, JI, J2, GAMMA2, GAMA3P, GAMA3, GAMAA, G	4	
	ALTAS, DELTA4, BETA1, BETA2, BETA3, D1, D2, ALTIN,		
	5A1, ALPHA2, IN, T2, T3, MU, MU1, RCP, PSIC, S1, S2, S4, S5, A1, A2, DBM13, DBS13, F	Ф (
	DIMENSION XXX EF238AX, EF12MAX, PNMAX, PN, AL PHEN, AL PHEX		
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	PHIT=(PHITOT+DELPHI)*22	-	
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	AA11.AA12.AA13 AA44 A445 A446 A446	5	
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		17	
	IF (X(4)*DPSI2.GE.O.) IPR=IP+AA22	æ	
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	(X(2) + DPHI2.GE.O.) IIR=	-20	
	(X(2)*DPHI2.LT.0.) IIR=	~	
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		-23	I
	IF (IPR.EQ.O.) WRITE (6,1)	24	
	AA51 = 1 PR	25	
	AASZ=2.+ONZ+AAZO/ABS(OMEGA)	26	
	ANDUSTRIAN TO	27	
	NACH DESCRIPTION OF THE PROPERTY OF THE PROPER	28	
	AA015113-1A715*N32412/AA444AA15*AA42*I1R*N31/(AA31*AA44)	67	
	AND COLAN COLORAGE AND COLORAGE	. c	
	AMMA HAR AMMA HAR AMMA HAR AMMA HAR AMMA HAR AMMA HAR HAR HAR HAR HAR HAR HAR HAR HAR HA	5	
	ABS0=AA1E+AA2E+AA46+12/AA44+AA15+AA32+AA42/(AA31+AA44)	32	
	DX(+)=X(+)	34	
	DX(3)=X(4)	35	
	DX(2)=(-AA56+X(2)+X(3)+AA67+V(3)+AA67	36	
	12+X(1)-PHIPR+2211/4859	37	
	DX(4)=(-AA21+X(4)+X(4)-AA52+X(4)-AA53+AA54-CTU/CAMAON	38	
	(AAS)	39	
	RETURN	40	
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	END COMMITTEE COURTS ZERO - SIMULATION TERMINATED)	44 An	
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1 LTA, PH 1TOT 2 P2, NP3, CAP 2 P2, NP3, CAP 2 P3, NP3, ELTA 4 LTA3, BELTA 5 A1, ALPHA2, 5 A1, ALPHA2, 6 A1,	, C, R, ALPHB, PI, , PHIPR.N31, N32 RB1, CAPRB2, RB2 1, J2, GARMA2, GA 1, J2, GARMA2, GA 1, BETA1, BETA2, M, BETA1, BETA2,	- c		
2P2, NP 2P2, NP 3H03, R 4LTA3, R 5A1, AL 5A1, AL 6238AX 6238AX CCCCC PHID= PHITO PHITO PHITO PHITO PHITO PHITO PHITO CALL	HITOT, PHIPR.N31, N32, GMEGA, GM2, RC1, PHI1C, TEST1, TEST2, NG1, NG2, N 3, CAPRB1, CAPRB2, RB2, RB3, THETA1, THETA2, R1, R2, R3, R4, RHO1, RHO2, R HOP, J1, J2, GAWMA2, GAMA3P, GALTA3, GAMA4P, GAMAPP, DELTA2, DE. DELTA4, BETA1, BETA2, BETA3, D1, D2, AL11N, AL1FIN, AL2FIN, ALPH PHA2, IN, T2, T3, MU, WU1, RCP, PSIC, S1, S2, S4, S5, A1, A2, DPHI2, DPSI2, F			
272, NP. 2103, R. 2103, R. 2103, R. 2224AX REAL DIREN DOI:100.100.100.100.100.100.100.100.100.100	J.CAPKBI, CAPKB2, RB2, RB3, THETA1, THETA2, R1, R2, R3, R4, RHO1, RHO2, R HOP, J1, J2, GANMA2, GAMA3P, GALTA3, GAMA4P, GAMAPP, DELTA2, DE DELTA4, BETA1, BETA2, BETA3, D1, D2, AL1 IN, AL1 FIN, AL2 IN, AL2FIN, ALPH PHA2, IN, T2, T3, MU, MU1, RCP, PSIC, S1, S2, S4, S5, A1, A2, DPHI2, DPSI2, F	. (7)		
4173, 541,411, 541,411, 6234AX REAL DIMEN COUTGO PHIDE PHITE PHITE PHITE	HUP.JI.JZ.GAMMAZ.GAMMAZP.GAMMAZP.GAMMAZP.GAMMAZP.GAMMAZPB.DELTAZ.DE DELTA4, BETA1, BETAZ, BETA3, D1, D2, AL1 IN, AL1 FIN, AL2 IN, AL2FIN, ALPH PHAZ, IN, T2, T3, MU, MU1, RCP, PSIC, S1, S2, S4, S5, A1, A2, DPHIZ, DPSIZ, F	•		
541,41,61,41,41,61,41,41,41,41,41,41,41,41,41,41,41,41,41	DECIMY, BEINT, BEINZ, BEINZ, BILS, ALTIN, ALTFIN, ALZIN, ALZFIN, ALPH PHAZ, IN, TZ, TZ, MU, MUL, RCP, PSIC, SI, SZ, SA, SS, AI, AZ, DPHIZ, DPSIZ, F FISHAY, FESSHAY, FESSHAY	•		
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Z. AA19	Z. AA19, AA20, AA21, AA22, AA23, AA26, AA26, AA27, AA28, AA29, AA31	2 5		
25AG0	AA37	22		
47	T.	29		
K) 41	(4)*DPSIZ:LT.0.) IPR=IP-AA22	24		
X	(2)*DPHI2.GE.O.) IIR=I1+MU*RHD1*(AA26+AA30)	. K		
Y) 4F	(X(2)+DPHI2:LT:0-)-E1R=E1-MU+RH01+(AA20+AA30)			
		22		
	PR.L. 0.) IPR.0.	28		
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A 654=(AACA CONTAINS OF C	31		
AA55=	A655=13-6415+N32+12/A8444-R444C+N34-1/A844-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-	32		
AA56=		33		
AA57=1	AA57=AA15+AA33+AA62+N34-(+&34+C+44+)	34		
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PS1=x(3)	[3]	37		
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DPS12	12 1 1 1 (d) 1 x (d) - 6 x 5 0 + 4 1 x 1 x 5 0 : 4 x 5 x 5 x 5 x 5 x 5 x 5 x 5 x 5 x 5 x	39		
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DPH12	(-4456ex(2)ex(2)=44E7ex(2)+44E0 exec.com/encode	14		
12+x(1	12+X(1)-PH(PR-27))/AASK	42		
		43		
CCOMPUT	COMPUTATION OF CONTACT FORCES	44		
		Ծ Հ		
FF23=(FF23=(13+DPH12+13+AA16)/AA15	9		
FF12=(FF12=(FF23*AA44+T2*AA43-12*N32*DPH12)/AA42	47	ı	
IF (Fi	IF (FF23.GT.FF23MAX) FF23MAX=FF23	2 4		
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3 I TAB	(6,6) T, PHID, X(2), PSID, X(4), PHITOT, FF12, FF23	55		

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•	3A29,AA30,AA31,AA32,AA33,AA34,AA35,AA36,AA37,AA38,AA39,AA41,AA	II M4		
	77 Mt M2 M2 M6 74	H		1
	1LTA, PHITOT, PHIPR, N31, N32, CMEGA, CM2, RC1, PHITC, TEST1, TEST2, NG1, NG2, M	X I		
	2P2. NP3. CAPREL, CAPREZ, REZ, REJ, THETAL, THETAZ, RI, RZ, R3, R4, RHO1, RHOZ	=		
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	6238AX, F12MAX, FF23MAX, FF12MAX, DMMAX, DM At DUEN At DUEN	٦.		
	REAL MI, M2, M3, MP, MU, MUI, N31, N32, 11			
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<u> </u>	MU=ABS(MU)*DPHI/ABS(DPHI)	_		
9	2	9 : T :		
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, 20	DELAL 2=DEI PHT = 27			
	DELAL1=DELAL2+R82/CAPR81	1		
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8	(ALPHAZ.LT.TEST2) S2			
	IF (ALPHA).EQ.TEST1) S1=0.			
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	TAR2=ABS((MP*(RA*(SIN(GAMAPP)-MUI*S9*COS(GAMAPP))+RCP*(SIN(Bet-Betz	K 52		
*	1)-MU1+55+COS(PSI+PSIC))))/DENOR1)			
,	AAG#ABS((MP#RCD#(S1%(DS1+DS1C)+BE11+S0F+)36+396+394+30+31-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-			
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9		86 20 20 20 20 20 20 20 20 20 20 20 20 20	
	AAB=ABS((MP*RCP*(SIN(PSI+PSIC)-MU1*S5*COS(PSI+PSIC)))/DENOM1) AA10=ABS((-MU*(1S2)*SIN(BETA2+THETA2)-(1.+MU*MU*S2)*COS(BETA2+TH		
59	AA11=ABS((SIN(GAMMA3)-MJ+COS(GAMMA3))/DENOM) AA13=ABS((-(1.+MJ+MJ+S2)+SIN(BETA2+THETA2)+MJ+(1S2)+COS(BETA2+TH	252	
	AA14=ABS((COS(GAMMA3)+MU+SIN(GAMMA3))/DENOM) AA15=RB3-MU+(S2+(D2-A2)+RHD3+(AA10+AA13)) AA16=MU+RHD3+(AA11+AA14)	4 66 1 67	
70	IF (ALPHR.EQ.ALPHEX) GO TO 7 AA1=ABS((-MU1*(S4+S5)*SIN(PSI+ALPHR)-(1S4*S5*MU1*MU1)*COS(PSI+AL 1PHR))/DEMOM1)	69	
75	AA5=ABS((MU1+(S4+S5)+COS(PSI+ALPHR)+(1S4+S5+MU1+MU1)+SIN(PSI+ALP 1HR)/DENOM1) AA9=ABS((MU1+S4+MU)+SIN(PSI+ALPHR+BETA2)+(1MU1+MU+S4)+COS(PSI+A	72 73	
	AA12=ABS(((1MU1*NU*S4)*SIN(PSI+ALPHR+BETA3)-(S4*MU1+MU)*COS(PSI+ 4ALPHR+BETA3)}/DENOM)	75	•
	AA17=ADNE-MU1*S4*BONE+MU*RHG3*(AA9+AA12) AA18=DONE-CONE*MU1*S4-RHOP*MU1*S5*(AA1+AA5) -G0.T0-8	78	
7	AA1=ABS((MU1*(-S4+S5)*SIN(PSI+ALPHR)+(1.+S4*S5*MU1*MU1)*COS(PSI+AL 1PHR))/DENOM1) -AA5=ABS((MU1*(S6-S4)*COS(PSI+A1PMD).(1.+S4.*SE*MU1*MU1).e************************************	H 81	
65	1HR))/DENOM1) AA9=ABS(((MU1*S4-MU)*SIN(PSI+ALPHR+BETA3)-(1.+MU1*MU*S4)*COS(PSI+A 4-PMR+BETA3)}/DENOM)	84 85 85	
8		H 88	
8	AA19=-DONE-RU1#54-RHOP+MU1#55*(AA1+AA5) - AA19=RHOP+MU1#55*(AA2+AA6) - AA20=RHOP+WU1*(AA3+AA7)	90	
36	AA22=RHOP*MU1*(AA4+AA8) AA23=ABS({(1- -MU*MU*S1)*SIN(BETA1-THETA1)*MU*(1:+S1)*COS(BETA1-THE 1TA1))/DENOM)	9.94	
100	AA24=ABS((M1*RC1*(COS(PHI1C+N31*PHIT)+MU*SIN(PHI1C+N31*PHIT))) 4)/DENOM) 4025=ABS((M1*RC1*(COS(PHI1C+N31*PHIT)+MU*SIN(PHI1C+N31*PHIT)))/DEN 10M) AA25=ABS((M1*RC1*(SOS(PHI1C+N31*PHIT)+MU*SIN(PHI1C+N31*PHIT)))/DEN	97.5 1 99.5 1 100	
105	1*(1.+S1)*	102 103 104	
0-1-	11T))/DENOM) 7A29=ABS((M1*RC1*(MU*COS(PHI1C+N3T*PHIT)=SIN(PHI1C+N31*PHIT)]/DEN- 1CM) AA30=ABS((M1*RC1*(MU*SIN(PHI1C+N31*PHIT)+COS(PHI1C+N31*PHIT)))/DEN- TOM)		
	CAPRB1-MU+S1+A1+I OM2+RHO1+MU+(AA2-I OM2/ABS(OMEGA)+2	H 110 H 111 H 112	
	AA34=KU+RHO1+(AA25+AA29)	11.5	ı

115 AA3 17A2 17A2 11A1 120 AA3 11A2	-1844 A 6 4 m a 2 4 mars
	AA39≒ABS(<u>((1,-MU÷MU÷S2)</u> +SIN(BETA2+THETA2)±MU*(1,±S2)÷COS(BETA2+THE H 116 1TA2))/DENOM)
AA40	I I I I
36+	BETA1-THETA1)-(1.+MU+MU+S1)+COS(BETA2+THE
	AA41=AB5((MU*COS(GAMMA2)+SIN(GAMMA2))/DENOM) H 125 AA42=RB2-MU*(R106) H 125 AA44=CAPPRO-MI*(52*A)
130 RETURN	Z I I

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SUBROUTINE ALFA	74/74 OPT=1 FTN 4.8+508 06/08/82 13.52.46 PAGE 1
1 0	B, THETA, CAPRO, RG, ALIN, ALFIN)
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	ALFIN-SQRI(CAPRO-CAPRB-CAPRB)/CAPRB 1 5 RETURN 1 7 END
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00 1 AMBDA = 02 020 N = 22				
TURNEST STATE				
.31850E-04 MZ = .32750E-05	113 = .26	26310E-05 MP =	.16400E-05	
.19670E-05 IZ40260E-07	13 = .21	.21250E-07 IP =	.19500E-07	
0576 RCP = 0.0000 RHOP =0227	7 RPM =30000	PHI1CD =-120,1340	PSICCD = 0.0000	PHID = 139, 0000
1485.				
MU1 = .10				
80.0 PSUBD2 =100.0				
NG2 = 36. NP2 = 9. NP3 =	ě			
.41214 CAPRP2 = .19039				
05796 RP3 x 04231				•
24.215 THETA2 = 27.326				
.25000 R2 = .31700 R3 = .30900 R	R4 = .30400			
-03075 - 1862 = .01500 - RIGG = .015	01500			
.37588 CAPRB216915 RB2	.05286 RB3 =	.03759		
.41425 CAPROZ = .19404 ROZ = .	.07670 RO3 *	.05580		
.00 BETA2D = 134.90 BE				
111147 GAMMAGO 155,09 GAMMA	140 186.34			
T = 0.00000 PHI = 139.00 PHIDDI = 0.0	5	H	0.00	00
-139:00 PHEDGT6:				90
= 2.5031 = 139.00	PN = .3305 41 G = .0222	PNPSI = .3305 PSID = 42.08	+07	'
= 2:5031 £12 = 9:6925 = 139.01 PHIDOT = 18.		PNPSI = 3306 PSID = 42.09	12E+07	
4 4	53 G-# 0229	PNPSI = .3309	= .1213E+07	
F23 = 2.5037 F12 = 9.6925 PHI = 139.02 PHIDDT = 30.			121.	
= 2.5047 F12 = 9:6924 = 139.03 PHIDDT = 36	¥ 50	PNPSI =:3316 PSID = 42.11	=	
F23 = 2.5048 F12 = 9.6925	Z.		= 31.30 PHILLIO	•
F23 = 2.5062 F12 = 9.6923	5 N		= 36	.04
.00004 PHI = 139.06 PHIDOI = 48.8	.82 G = .0220	PSID = 42.13	PSIDOT = 41.71 PHITOT =	90*
× 139.07	5		Orniz * .121/E+U/	

60.	11.		13	in in	.17	ç		.22	.25	ě		.32	.35	. (.42	1	P F	125	ŭ	3	.59	64		69.	.74	30		.85	96.		.	1.02	1.08)	1.14	1.21		1.28	1.35	1.42
PHITOT .	PHITOT =	7	7	PHITOT .	PHITOT =	7 DHITGT =		PHITOT *	PHITOT =	7 Burtor -		PHITOT .	PHITOT =	7	7	PHITOT =	Outtot -	7	PHITOT =	7 PHITOT *	7	PH1101 .	7 PHITOT		PHITOT =	PHITOT =	7		PH1107 =	PHITOT .	7 - 041707	~	PHITOT =	PHITOT =	7	PHITOT =	PHITOT .		# 101 #4 7	PHITOT	PHITOT =
52.16	57.41	. 1225E+07	12245+07	67.96	73.26	.1228E+0	. 1235E+07	83.94	89.31	. 1242E+07	. 1239E+07	100.14	105.59	9	. 1256E+0		1253E+37	. 1264E+07	-127.76 -	. 1261E+07 133.39	1273E+07	139.06	1269E+0	. 1282E+07	150.53		. 1292E+07	.1288E+07	168.09	174.04	. 1298E+07	. 1314E+0	186.11	192.23	. 1306E+07	198.37 1301E±07	204.48	. 1282E+07	210.58 .1276E+07	216.65	222.72
PSIDOT =	PSIDOT =	DPHI2 =	_ rs1001_= _ OPH12 =	PSIDOT =	PSIDOT =	DPHI2 = PSIONT =	DPHI2 =	PSIDOT =	PSIDOT =	OPHI2 =	DPHI2 =	PSIDOT .	PSIDOT =	DPHI2 =	DPHI2 =	PSIDOT =	PSIDOT -	DPH12 =	- PSIDOT -	DPHI2 = PSIDOT =	DPHI2	PSIDOT =	- PSIDAT-	DPHI2 =	PSIDOT =	PSIDOT	- PSIONT -	DPHI2 =	PSIDOT =	PSIDOT .	DPH12 =	DPHI2 =	PSIDOT =	PSIDOT =	DPHI2 =	- PSIDOT-	PSIDOT =	— DPHI2 ±-	DPHI2 =	PSICOT -	PSIDOT .
42.16	= 42.17	= .3355		11 1		= 42.25	n	= 42.27	4	= .3424 = 42.32	= .3	42.35	= 42.38	= .3459 - 42.41	,	= 42.45	*	"	- 42.52	= .3530 = 42.55	3564	= 42.59	42.63	н	= 42.67	= 42.72	42,76	#	= 42.81	11	42.91		42.96	H H	= .3795	= 43.07	"	- 3800	H 11	43.25	. 4
.0219 PSID	8	1355 PNPS1	-	.0217 PSID	9	1379 PNPSI 0215 PSID	-	0214 PSID 1402 PNPSI	6	424 PNPSI	•	0211 PSID 453 PNPSI		1459 PNPSI		.0207 PSID		.		3530 PNPSI .0202 PSID	54 PNPSI	0	0199 PSIB	-	0197 PSID		0193 PSID	a.	015d 1610	ð	0187 PSID		.0184 PSID	~	_	3808 PSID	_	3800 PNPSI	, .	3804 PSID	•
G = .021	2	PN = .3355	PN = N	" " " " " Z		FN =	PN = .3	. G = .3	= 5	E	F		 	PN = .3	PN = .3	וו טופ	= 5	# Md		FN # .3530 5 G = .0202	C NA	. c		P. = .3	P. C.	 		E. = N4		# (5)			# # 5 # 20	: U		,	"		Z Z	DN = 3804	
31 = 61.00	1. 67. 1	= 9.6924	9.6926	01 = 79.34 = 9.6924	or = 85.48	= 9.6927 XOT. = 91.63	9.6925	= 9.6929	ot = 103.96	= 9.6925 XOT = 110.15	9.6930	XOT = 116.35 ≅ 9.6927	OT = 122.57	= 9.6932 3T = 128.81	9.6928	PHIDOT = 135.07 F12 = 9 6934	JT = 141.34	.6930	JI.= 147-54 - 0 6036	= 9.0936 101 = 153.99	9.6832	101 = 160.29 - 9 6839	JT. =- 166.65	9.6935	F12 9.6943	oot = 179.43	PHIDOT-= 185.86	9.6947	- 9.6942	[= 198.7	1 = 205.30	9.6946	301 = 211.82 = 9.6957	01 = 218.39	9.6984	= 9.6995	DT = 231.39	= 9.7052 T	9.7064	PHILDS: 244-T	or = 250.4
139.09 PHIDOT		F12		PHID F12	H	EE			H	PHI	F12	F12	PHI	2.5250 F12 = 9 139.39 PHIDGE						E		PH10	NA NA	2.5466 F12		_		F12	F12	PHIC	-		2.5691 - F12 = 0	PHI	F12			2.3245 F12 # 140.28 PHIDOT		40.35 FHIDE	_
PHI = 139 F23 = 2.5	i n	F23 = 2.5105 PHI = 139.13	1 11	F23 = 2.5134	# (H 1	F23 = 2.5	Ħ	F43 = 2.5205 PHI = 139.28		F23 = 2.5247	•	PHI x 139		F23 = 2-5298	Ħ	F23 = 2.5347	16.851 - 144.51	H H		PHI = 139.59	4	F23 = 2.5	1	PHI = 139.74		F23 = 2.5539		PHI = 139.90	į	H	F23 = 2.5	н	F23 = 2.3204	1 11	PHI = 140	H H		F23 = 2.3	#
	. 00005	90000		90000	.00007	70000		9000	90000	60000			.00010	11000		. 1000	.00012			. 00013	-	. 00013	41000	71000		.00015	-51000	21000	-	.00016	. 00016		- 1	.00017	44000		.00018	.00019		n	. 00020
# 	-	-	! .	-	-	1.	•	- 1	-	1	•		-	1	•	-	-	•		-	•			•	- 1	-	-	-	- 1		*	•	-	-		•	-	-		•	-

1.45	: ;	1.49	1.53			1.60	25		189-7	•	1.72	1.78		8	1.84		.	\$:	1.96	2.00		2:04	2.09		?	2:17	2.23	(2.26	2.3	36.		7.40	2.45	•	•.	2.54	20.0	K C.7	2.64	
PHITOL		PH1101 =	PHITOT .	7	20.00	PH1101 =	PHI TOT .		PHITOT	, anstant		PHITOT =			PHITOT .		PHI 701 -	PHITOT		PHITOT .	PHITOT .		PHITOT	PHITOT .			PHITOT =	FHITOT =		- Include - 2	FHITOT -	7 PHITOT =		PHITOT *	PHITOT =	7 Puttot -		PHITOT .	Distrat		PHITOT .	7
.1250E+07	.1152E+07	228.41 1156F±07	•	.1169E+07	- 234-13 - 1166F+07	237.02	111/6E+0 939-92	.1172E+07	242.85	.1182E+07	- 11179F+07	248.75	.1189E+07		254.72	1198E+07	11925407	- 260.77	-	263.83	266.90	. 1209E+07	- 270.00	273.12	-: 1216E+07	. 1212E+07	- 279.41	282.59	- 1219E+07	285.79 .1230E+07	289.01	.1225E+07 292.26	. 1237E+07	295.53 .1232E+07	298.81	.1224E+07	. 1219E+07	305.37	.1205E+07	. 1200€+€7	311.91	.1185E+07
DPH12 = PSIDOT-=	DPH12 =	- DPHIO	PSIDOT =	DPHI2 =	DPH12 =	PSIDOT =	PSIDOT *	DPHI2 =	- PSIDOT.	DPH12 =	- DPHI2	PSIDOT .	DPHI2 =	DPHI2 =	PSIU01 =	- OPH12 =	PS100;	PSIDOT =	DPHI2 =	FSIDUT =	PSIDOT =	DPHI2 =	- PSIDOT -	PSIDOT =	DSTOOT	DPH12 =	PSIDOT :≡ DPHT2 =	PSIDOT =	- DPH12 =-	DPHI2 *	PSIDOT =	PSIDOT =	DPHI2 =	PS1301 =	PSIDOT =	DSIDET =	DPHI2	PSIDOT =	PSIDOT -	DPHI2 =	PSIDOT .	CPHI2 =
.3819 -43.35 -	.3681	3688	43.41	3718		43.48	43.51	. 3763	43.55	43.58	-3803-	43.62	.3835	.3844	43.69	3877-	3886	-43:76-	.3920	43.80	43.84	.3965	39.88	43.92	43.96	.4020	44.00	44.04	4068	.4106	44.12	44.16	.4157	4168	44.25	44.29	4184	44.33	44.38	.4189	44.42	44 47
PNPSI =	= ISANA	PNPSI	PSIO =	* ISANG	PhPSI =	PSIG =	PSID =	PNPSI =	+ 015d	* ISANA	- ISdNd-	PSID =	PNPSI .	PNPSI =	PS10 =	+ 1523 - 5132	PNPSI =	PSID	PNPSI *	PSID =	PSIO =	PNPSI =	= DISA	PSID =	FINEST	PNPSI =	PASTO =	PSID =	FISHNA	PNPSI =	PSID =	PSIO =	PNPSI =	PNPSI =	# 015d	PS10 =	PAPSI	PSID =	PSTD #	PNPSI =	PSID =	PSIO =
3819	3681	3688	.0165	3718	.3725	= .0163	0161	.3763	* 0350	5,5735 158	3803	0157	3835	3844	€ .0154	- 3877	.3886	-19191	. 3920	0149	= .0147	.3965	3974	0144	- 0142	.4020	4058	= .0139	.4068	.4106	± .0136	± .0134	4157	.4168	•	0129	4184	4170	- 0125	.4189	. 0123	. 4163
37	5.26 PN =	!	. 16	5 MA 6		4.99 G	7.92 G	_	\$	3.79 G	. 1	73	# 4 P	•	2.67 G	F. F.	•	8:63 G		1.52 6	4.63 G		5 5 5 4 G	S	3.68 G	P	9 PN = 0	.75	8 PN E 2.81 G	•	5:86 - G	9.	# PA 10 0	•	5.09 G	15		1.18 5. PN G	19	_	7.18	14
F12 = 9.7136 PHIDOL = 253	F12 = 9.7315 PH1007 = 256	6	ĕ	F12 = 9./319 PHIDGT = 262		PHIDOT = 264	24100T = 267	F12 = 9.7330		, <u>5</u>	u	PHIDOT = 276	FIZ = 9.7332 PHIDOI279	F12 = 9.7339	PHIDOT = 282	#12-4-9-1337 DHIDOT - 285		ğ	o,	PHIOUI = 291 F19 = 4 7350	PHI DOT = 294		rnibol = 237 F12 = 9.7355	PHIDOT = 300	PHIDOT = 303	6	F12 = 9.7359	PHIDOT = 309	F12 = 9.7368 PHIDOT = 312	6	PHIDOT = 315	PHI DOT = 318	F12 = 9.7373 PHIDOT = 322	6	PHIDOT = 325 612 = 0 7413	PHIDOT = 328		PHIDOT = 331 F12 = 9 7465	젖		F15 = 0 7617	ַ אַ
	= 2.5807	z 2.581		= 2.383/ = 140.56		= 140.60 = 2.590B				= 140.72	2.5965	= 140.76	140.80	_	* 140.84	140 AB	= 2.6075			= 140.96	141.00	= 2.6185			= 141.13		= 2.6306		= 141.26	_	= 141.31				= 141.45 = 2.3861		366	= :41 .54 = 2.3891	41.59		2.3922	
	¥23			L CS		PHI			2			E			H			1	F23			F23	F23		EE	F23			E		F23		F23 :	İ	F. E. 23		F23	. E23			123	
00020	. 00020		.05021	00021		.00021	.00021	600		. 00022		. 00022	. 00022		.00023	.00023		-00023	66000	. 6665	. 00024	96000		.00024	.00024		3002.	.00025	.00025		. 00025	.00026	.00026		97000	.00026		. 60027	. 60027		. 20027	. 00027
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5	2.13	2.78	2.83	9 6	3	2.93	2.98		3.04	3.09	•		3.19	3.24		3.30	3.35		3.40	2 46		3.51	3.57	;	3.62	3.68	3.74	•	-3:79	3.85	10 6	5	76:8	4.02		D	4:14	;	4.20	4.23	!	4.26	4.29
purtor -	-07	PHITOT =	PHITOT =	+07 PH1707 =		PHITOT =	PHITOT .	+07	PH1101 =	PH1101 =	FO7	+07	PHITOT .	PHITOT =		PHITOL	PHITOT *	107	PHITOT =	PHI TOT =	+07	PHITOT =	PHITOT =	101	PHETOT	PHITOT .	PHITOT =		PH 1101	PH1101 =	PHITOT =	-07	PHITOF-	PHITOT *			- PHITOT =		= 1011E4	PHITOT =		- PHI 101 - 1	PHITOT .
T = 210 At	- 1165E+07		= .1160E+U/ [_=_324,71	= .1066E+07 I = 327.74	,	T = 330.79	4		- 335.98 1080E+02		= .1075E+07 [= 343.26	= .1087E	e • ,	T = 349.64		T-=- 352.86	H	1	T = 359.40 = 1004E107	1362.71			T = 369.41	H	F - 372,80 -	. W		= 11	F-=-383.14	· m	=:1119E+07 [= 390.15		= 393.66 	r = 397.16	= .1075E+07	r •	4	= .10			9531E+06	= 410.89	
.51 PSIDO	į	1.56 PSIDOT	1	4043 DPHI2 4.65 PSIDGT	Ī	4.70 PSIDGT 4099 DPHI2	Ì	4110 DPHI2			4.89 PSIDOT		4.94 PSIDO	1		5.04 PSIDOT		338	5.14 PSIDOT 4352 DPH12			19.25 PSIU01 4417 DPHT9		_	5.35 - PSIDO1 4484 DPHI2		4537DPHI2- 5.46 PSIDOT		4610 DPHT2		4626 DPH12 5.62 PSIDO1		5.68 PSIDOT 4634 DPH12		4627 OPHI2 5.79 PSION	_	1	4636 DPHI2	-		4482 DPH12	4489 DPHI2	
PS10 = 44	PNPS1 =	PSIO = 44	PSID = 44	PNPSI = .4		PSID = 44	4	PNPSI = .4	PNPSI =4	PSID = 44			PSID = 44	4 4	H	PSID = 45	. 4		PNPSI = 45	4 = 4	# (PAIN = UISA	"	11	PNPSI = 4		PSID = 45	#	PNPSI = .4	#	PSID = 45		PNPSI = 45	#	PNPSI =4		1	PNPSI = .45		#	PNPSI = .4		PSID = 46
G = .0120	4188	G * .0118 * .4199	G = .0116	= .4043 G = .0114	1	€ = .0112 = .4099	0 = 5	G = .010A	4	G = .0106	2 7	H C	4228	G = .0100	= .4276	= .4289	9600. = 5	= 4338 G = 0004	. 4.	G = .0092	6 × 0080	4417	G = .0087	4469	= .4484	G = .0083	G = .0081	= .4552 6 - 0020	= .4610	G = .0076	G = .0074	4	= .4634	G = .0070	G = .0067	€ .4643	- 000 5	G = .0063	4652	G = .0062	4482	= .4489	6500. = 5
= 343.08	1.7571	= 345.99 9.7581 PN	8	9.7759 PN = 351.33	υ,	3.7768	= 356.65	= 359,33	1777.	= 362.01 9.7788 PN	= 364.70	9.7787 PN	798	Ξ	9.7797 PN	9.7809	.55	378 20	9.7820	= 381.02	9.7820 PN	•	= 386.52	9.7832 PN	9.7845	= 392.05	8	9.7858 PN	.7856	= 400.40	18	9.7923 PN		۳	9.7990 PN = 411.33		= 413.98···	= 416.60		8	3.0250 PN	9.8257 PN	= 420.18
	F12 =	F12 =	PHIDO	PH100	F12 =	F12 =	PHIDOI	PH1001	£12 =	F12 =	PHIDOI	F12 =	E12.E	PHIDOI	F12 =	F12 =	PH1001	40 PHIDOT - 378	F12 =	PHIDO	F12 =	£12	PHI	F12 = DHIDOT	F12 =	PHIDO	PHIDO	F12 =	F12 =	PHIDOT	PH1001	F12 =	F12 =	PHIDOT	PH1001	F12 =	FH1001	PHIDOT	Ī	PH1001	E TOUR	F12 =	29 PHIDOT
PHI = 141.73	$\frac{F23}{641} = 2.39$	= 2.3	141	H H	-173 = 2.65	şê.	F23 = 2 6583	1 11	*	F11 = 142.09		F23 = 2.6725 PHI = 142.19	P		PHI = 142 20		PHI = 142.35	4 #	F23 = 2.6886	1	F23 = 2.6958 PHI = 142.51	*		F23 = 2.7039	W	PHI = 142.	-	F23 = 2.7130 -PHI = 142.79	Ħ	PHI = 142.85		F23 = 2.45	2.4	PHI = 143.02	!	n .	F71 = 143.14		ji Ca	PHI = 143.23	,	Ħ	PHI = 143.
00028	90000		* .00028	. 00028	.00029		• .00029	00029	00000	62000 =	a 000030	00030		. 00030	000030		00031	00031		* 00031	00031		. 00032	- 00032	,	00032	00032	00033		00033	00033	- 00033		00034	00034		*******	* .00034		00035	- 00035		00035
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0045 PHI = 211.69 PHIDDT = 612.91 PSI = 315.86 FF12 = 9.427 FF23 = 2.444 0046 PHI = 211.74 PHIDDT = 620.77 PSI = 315.92 VS= -71.706 WOTION 0046 PHI = 211.782 PHIDDT = 113.856 PSI = 315.922 VS= -13.152 PHI = 211.782 PHIDDT = 113.866 G =6145 PS 0046 PHI = 211.78 PHIDDT = 113.86 G =6145 PS 0047 PHI = 211.85 PHIDDT = 125.13 G =0113 PS 0048 PHI = 211.93 PHIDDT = 136.50 G =0114 PS F23 = 2.4002 F12 = 9.7742 PN = .3863 PNP 0049 PHI = 211.93 PHIDDT = 147.94 G =0114 PS F23 = 2.4002 F12 = 9.7742 PN = .3863 PNP 0049 PHI = 212.01 PHIDDT = 147.94 G =0108 PS PSI = 2.4001 PHIDDT = 147.94 G =0108 PS	= 459.50 PH]	PHI TOT = 7.20	
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BLICANT ALONG THE CONTRACT OF	1	-163.12 PHITOT =	1.95
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5.45 PSIDOT	ĺ	_		3617 OPHI2 5.28 PSTO			.16 PSIDO1		5.10 PSIDO1 3636 DPH12		1660 DPH12			3670 DPH12	- 64			4.70 PSIDOT		_	1	3667 DPH12			3625 DPHI2		SAS PSIDOT	.22 PSID01	1359 DPH12	353 DPH12	E	1.15 PSIDOT	1355 OPHI2	363 DPHT2		357 DPHI2	. w	Ī	1359 DP1112	367 DPHT2	_	3361 DPH12		
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= 170.45 9.7878 PN	= 176.02	9.7856	= 181.67	= 187.41	9.7817	= 193.23 9 7801 BM	= 199.16	9.7775 PN	9.7757	= 211.30	3.7729 PM	9.7710	r = 223.86 a 757a an	= 230.30	9.7659		9.7625 PN	9.7603	= 250.38	9.7566	= 257.32	9.7543 PN = 264.41	9.7544	= 271.56	= 278.75	9.7546	. = 285.99 9.7521 PN	= 287.71	9.7713 PN	9.7707	. = 291.12 o 760e bu	= 292.85	9.7690 PN	.7679	± 296.32	5.7073·······	9.7661	≈ 299:83 o zere	301.60	9:7644	= 303.37	9.7637 PN	9.7625	306.96
.19 PHIDOT 580 F12 =	8	F12 =	HE	۱ 8	F12 =	F 25 2	PHID	6728 F12 =	F12.	_		F12 =	PHIDO	PHID	F12 =	8	F12 =	E12	\mathbf{a}			.11 PHIDOT	E12	H	-		.35 PHIDOT		.6899 F12 = 13.39 PHIDOI		13.41 PHIDOT	.43 PHIC	912 F12 = 9		PHID	3.50 PHIDOT		3.52—PHIDGT		Ī	PHIC	.56 F12 =		.61 PHIDOT
5 6	2	. 2	F12 = 212	, , ,	2	F23 = 2.6	= 21	F23 = 2.67	2 2	PHI = 212	, E	. 2	PHI = 212.	= 212	= 2.6	_		H 2	# 21	e i	50.5 = 213.04	PHI = 213.11	M	PHI = 213.19	1		PHI = 213.	H	F23 = 2.68	11	PHI = 213.	#	F23 = 2.6	= 2.6	PHI = 213	# 213	6	F22 - 213	 	- 2	۳.	F23 = 2.6	= 2.6	PHI = 213
00051	00051		00052	00052	- 00063	2000	00053	A2000	- l	00054	00055		- 00055	₹ .00056		95000	00057	ļ	.00057		BCOMO	00058		00059	- 000059		09000 -	00060	09000		● .00060	• • • • • • • • • • • • • • • • • • • •	09000		• .00060	• .00060		19000	= .00061		. 00061	19000		. 00061
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~			3469 .0024 3447 p	ļ ·]	3415 F					.0000 3021	= 45.41	= 45.36	= 45.30	= 45.24	= 45.18 = 45.13	'		- 44.90
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= 38 0-714	PHIDOT = 38 F12 = 10.710	5-5	40 "	9 4 9	F12 = 9.766 PHIDOT = 41 F12 = 9.759	PHIDOT 42 F12 = 9.757 PHIDOT = 42	F129,750 PHIDOT = 43 F12 = 9.748	PHIDOF = 43 F12 = 9.741 PHIDOT = 44	H Q H	Ö + Ö	=10.594 DOT = 46 =10.594	- 46	ų jų į	, <u>"</u>	M	16 16	, * ,i		FF23 = 2.
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.05147	.05147	.05147	.05148	.05148	.05148	.05149	.05149	.05150	.05150	.05150	.05151	MOT 10N	.05151	.05151	.05151	.05152	.05152	.05153	.05153
- AFAA - Area -	= .05147 PH1 = 214.26 PH1001 = 382.70 G = .0032 PSID = 31 \pm 2.9621 E12 = 10.7144 PN = 3459 PNPSI =	= .05147 PH1 = 214.26 PH100T = 382.70 G =0032 PSID = 31 = .05147 PH1 = 214.31 PH100T = 387.25 G =0030 PSID = 31 F23 = 2.9679 F12 =10.7106 PN = .3478 PWPSI = .	= .05147 PH1 = 214.26 PH100T = 382.70 G =0032 PSID = 31 = .05147 PH1 = 214.31 FH100T = 387.25 G =0030 PSID = 31 = .05147 PH1 = 214.37 PH100T = 391.85 G =0028 PSID = 31 = .05147 PH1 = 214.37 PH100T = 391.85 G =0028 PSID = 31 = .05147 PH1 = 214.43 PH100T = 396.52 G =0026 PSID = 31	.05147 PH1 = 214.26 PH1001 = 382.70 G =0032 PSID = 31 .05147 PH1 = 214.31 PH1001 = 387.25 G =0030 PSID = 31 .05147 PH1 = 2.9679 F12 =10.7106 PN = .3476 PNPSI = . .05147 PH1 = 214.43 PH1001 = 391.85 G =0028 PSID = 31 .05147 PHI = 214.43 PH1001 = 396.52 G =0026 PSID = 31 .05148 PHI = 214.48 PH1001 = 396.52 G =0026 PSID = 31 .05148 PHI = 214.48 PH1001 = 401.23 G =0024 PSID = 31 .05148 PHI = 214.48 PH1001 = 401.23 G =0024 PSID = 31	.05147 PH1 = 214.26 PH100T = 382.70 G =0032 PSID = 31 .05147 PH1 = 214.25 PH100T = 382.70 G =0032 PNPSI = .05147 PH1 = 214.31 PH100T = 387.25 G =0030 PSID = 31 .05147 PH1 = 214.37 PH100T = 391.85 G =0028 PNPSI = .05147 PH1 = 214.43 PH100T = 396.52 G =0026 PSID = 31 .05147 PH1 = 214.43 PH100T = 396.52 G =0026 PSID = 31 .05148 PH1 = 214.48 PH100T = 406.00 G =0024 PSID = 31 .05148 PH1 = 214.49 PH100T = 406.00 G =0024 PSID = 31 .05148 PH1 = 214.60 PH100T = 406.00 G =0023 PSID = 31 .05148 PH1 = 214.54 PH100T = 406.00 G =0023 PSID = 31 .05148 PH1 = 214.54 PH100T = 406.00 G =0023 PSID = 31 .05148 PH1 = 214.60 PH100T = 410.81 G =0021 PSID = 31	.05147 PHI = 214.26 PHIDDT = 382.70 G =0032 PSID = 31 .05147 PHI = 214.31 PHIDDT = 382.70 G =0030 PSID = 31 .05147 PHI = 214.31 PHIDDT = 381.85 G =0030 PSID = 31 .05147 PHI = 214.37 PHIDDT = 391.85 G =0028 PSID = 31 .05147 PHI = 214.43 PHIDDT = 391.85 G =0026 PSID = 31 .05147 PHI = 214.43 PHIDDT = 396.52 G =0026 PSID = 31 .05148 PHI = 214.48 PHIDDT = 401.23 G =0024 PSID = 31 .05148 PHI = 214.54 PHIDDT = 401.23 G =0024 PSID = 31 .05148 PHI = 214.54 PHIDDT = 401.81 G =0024 PSID = 31 .05148 PHI = 214.64 PHIDDT = 410.81 G =0021 PSID = 31 .05148 PHI = 214.66 PHIDDT = 410.81 G =0021 PSID = 31 .05148 PHI = 214.66 PHIDDT = 410.81 G =0021 PSID = 31 .05148 PHI = 214.66 PHIDDT = 410.81 G =0019 PSID = 31 .05148 PHI = 214.66 PHIDDT = 410.81 G =0019 PSID = 31 .05148 PHI = 214.66 PHIDDT = 415.89 G =0019 PSID = 31 .05148 PHI = 214.66 PHIDDT = 415.89 F = .3437 PNPSI = .	PHI = 214.26 PHIDDT = 342.70 G =0032 PSID = 31 F23 = 2.9621 F12 =10.7144 PN = .3459 PNPSI = . PHI = 214.31 PHIDDT = 391.85 G =0030 PSID = 31 F23 = 2.9666 F12 =10.7106 PN = .3476 PNPSI = . PHI = 214.43 PHIDDT = 396.52 G =0026 PSID = 31 F23 = 2.9665 F12 =10.7084 PN = .3458 PNPSI = . PHI = 214.48 PHIDDT = 396.52 G =0026 PSID = 31 F23 = 2.9655 F12 = 9.7768 PN = .3469 PNPSI = . PHI = 214.48 PHIDDT = 406.00 G =0024 PSID = 31 F23 = 2.9663 F12 = 9.7683 PN = .3453 PNPSI = . PHI = 214.60 PHIDDT = 410.81 G =0021 PSID = 31 F23 = 2.9588 F12 = 9.7661 PN = .3453 PNPSI = . PHI = 214.66 PHIDDT = 410.81 G =0021 PSID = 31 F23 = 2.9569 F12 = 9.7661 PN = .3437 PNPSI = . PHI = 214.72 PHIDDT = 420.61 G =0019 PSID = 31 F23 = 2.9535 F12 = 9.7555 PN = .3445 PNPSI = . PHI = 214.72 PHIDDT = 420.61 G =0015 PSID = 31 F23 = 2.9535 F12 = 9.7573 PN = .3445 PNPSI = .	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APPENDIX B

MANAGES SECTION OF SECTIONS SECTIONS OF

DYNAMICS OF ROTOR DRIVEN S&A MECHANISM WITH A TWO-PASS

INVOLUTE GEAR TRAIN AND A VERGE (PLATE PALLET) RUNAWAY ESCAPEMENT

Description of Systems and Outline of Derivations

This appendix gives derivations for the complete mathematical model of an S&A mechanism, consisting of a spin driven rotor, a two pass, step-up involute gear train and a verge-type runaway escapement. Figures B-1 and B-2 show the two types of configurations to which this model may be adapted.

This work uses geometric and kinematic equations for the verge escapement developed in appendixes C through F. In addition, the dynamics of fuze gear trains is based on the work reported in reference B-1. As in previous efforts of Lowen and Tepper on the pin pallet escapement (refs B-1 and B-2), the following three motion regimes of the mechanism are considered:

Coupled Motion

The tip of the escape wheel is in constant contact with either the entrance or the exit working surface of the verge while it is driven by the rotor (gear no. 1) through the gear and pinion set no. 2. The entrance—as well as the exit—coupled motion differential equations, which differ from each other to some extent, are expressed in the escape wheel variable ϕ and are obtained by combining the solutions to the Newtonian force and moment expressions for the individual mechanism components.

Free Motion

The pallet and the escape wheel, gear train, rotor systems move independently of each other in this phase of the motion. Thus, there results one differential equation for the pallet in the variable ψ , and another one for the combined system in the escape wheel variable ϕ .

Impact

The formulation of the impact regime is similar to that shown in reference B-1 for the pin pallet escapement. Again, the moment of inertia of the escape wheel and pinion no. 3 also contains the referred mass properties of the rotor and gear and pinion set no. 2. This impact simulation is based on the classical angular impulse - momentum model, where a coefficient of restitution is used to account for the energy losses. Since it is assumed that the effect of the impact force between the escape wheel and the pallet is much greater than the effects due to the driving torque of the rotor and the various retarding torques due to friction, the latter are disregarded.

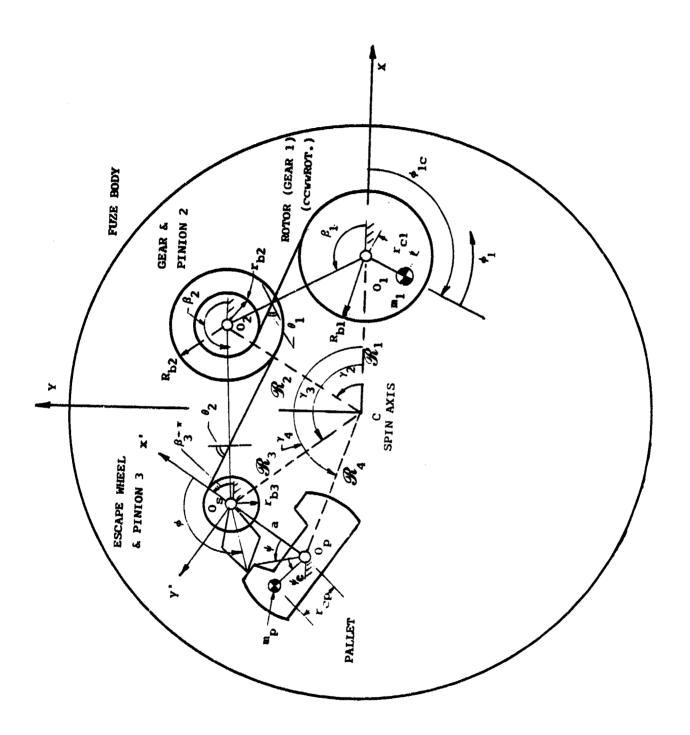


Figure B-1. Rotor driven S&A device, configuration no. \mathbf{l}

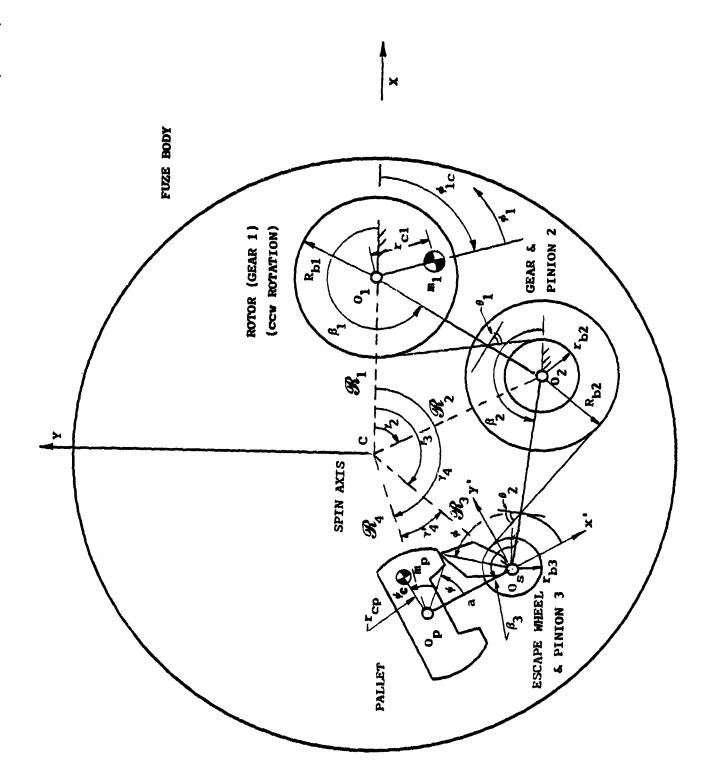


Figure B-2. Rotor driven S&A device, configuration no. 2

Treatment of Friction Forces

The influence of friction forces is considered both in the coupled and the free motion regimes. There is friction at the interface between the escape wheel and the pallet during coupled motion, and there is friction between the teeth of the gears at all of the pivots during both of these types of motions. As in reference B-1, the individual pivot friction torques are obtained by the algebraic addition of the two friction moments due to the x and y components of the normal bearing forces, rather than by the direct use of the resulting normal forces. This conservative approach to friction is necessary to avoid the difficulties which the presence of a square root term introduces into the solutions of the various differential equations.

Appendix E of reference B-1 proves that a change in sign of the coefficient of friction, both in the coupled and the free motion simulations of the gear train, including the rotor and the escape wheel (with the exception of the escape wheel-pallet interface), is sufficient to account for the changed directions of the friction forces encountered during a possible motion reversal of the gear train following an impact.

Geometry of Fuze Body Configurations

The following section contains derivations of expressions for the various fuze body angles associated with the pivot holes of both configurations of the two pass, step-up gear mechanisms. In addition, relationships between the unit vectors of the body-fixed X-Y and x'-y' systems (figs. B-1 and B-2) are given to describe certain escapement forces in terms of the primed coordinate system. (Note that the x'-axis is oriented along the escapement center-line $O_p - O_s$.) Finally a signum function is introduced so that common programming expressions for both configurations can be devised.

Fuze Body Configuration No. 1

The following relationships for the angles γ_i , δ_i , and β_i for configuration no. 1 are indicated by figure B-3:

$$\gamma_2 = \cos^{-1} \left[\frac{\frac{2}{1} + \frac{2}{2} - (R_{p1} + r_{p2})^2}{2R_1 R_2} \right]$$
 (B1)

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Figure B-3. Fuze body configuration no. 1

Further.

$$\gamma_3' = \cos^{-1} \left[\frac{R_2^2 + R_3^2 - (R_{p2} + r_{p3})^2}{2R_p R_3} \right]$$
 (B2)

Finally,

$$\gamma_3 = \gamma_2 + \gamma_3^{'} \tag{B3}$$

The angle γ_4^{\prime} is given by:

$$\gamma_{4}' = \cos^{-1} \left[\frac{R_{3}^{2} + R_{4}^{2} - a^{2}}{2R_{3}R_{4}} \right]$$
 (B4)

Then,

$$\gamma_4 = \gamma_3 + \gamma_4' \tag{B5}$$

Angles δ_1 . The angles δ_2 and δ_3 , as shown in figure B-3, become:

$$\delta_2 = \cos^{-1} \left[\frac{(R_{p1} + r_{p2})^2 + R_1^2 - R_2^2}{2R_1 (R_{p1} + r_{p2})} \right]$$
 (B6)

and

$$\delta_3 = \cos^{-1} \left[\frac{(R_{p2} + r_{p3})^2 + 6\ell_2^2 - 6\ell_3^2}{2R_2(R_{p2} + r_{p3})} \right]$$
 (B7)

The angle δ_4 must take the pivot-to-pivot distance a of the escapement into account. Therefore,

$$\delta_4 = \cos^{-1} \left[\frac{a^2 + R_3^2 - R_4^2}{2aR_3} \right]$$
 (B8)

Angles β_1 . The angles β_1 and β_2 become respectively:

$$\beta_1 = \pi - \delta_2 \tag{B9}$$

and

$$\beta_2 = \gamma_2 + \pi - \delta_3 \tag{B10}$$

The angle β_{q} is found with the help of angle δ_{k} of equation B8. Then,

$$\beta_3 = \gamma_3 + \pi - \delta_4 \tag{B11}$$

Figure B-3 shows the angle γ^* between the positive x*-axis and the unit vector \overline{n}_Δ . It is given by:

$$\gamma_{\rm D}^* = \pi - \varepsilon \tag{B12}$$

where ϵ is obtained with the help of equations B4 and B8:

$$\varepsilon = \pi - \delta_4 - \gamma_4^*, \tag{B13}$$

and therefore:

$$\gamma_p^* = \delta_4 + \gamma_4^* \tag{B14}$$

The unit vector \mathbf{n}_4 is expressed in terms of the primed coordinate system as follows:

$$\vec{n}_4 = \cos \gamma_p^{\dagger} \vec{i}^{\dagger} + \sin \gamma_p^{\dagger} \vec{j}^{\dagger}$$
 (B15)

Further, the unit vectors \mathbf{I}^* and \mathbf{J}^* , when expressed in the X-Y system, become:

$$\vec{i}$$
' = cos $(\beta_3 - \pi)\vec{i}$ + sin $(\beta_3 - \pi)\vec{j}$

or

$$\vec{i}' = -\cos\beta_3 \vec{i} - \sin\beta_3 \vec{j} \tag{B16}$$

and

$$\vec{j}' = \vec{k}' \times \vec{1}' = \sin\beta_3 \vec{i} - \cos\beta_3 \vec{j}$$
 (B17)

Fuze Body Configuration No. 2

The angles $\gamma_{\bf i},~\delta_{\bf i},~$ and $\beta_{\bf i}$ of configuration no. 2 are defined in figure B-4.

Angles γ_1 . Since these angles are defined in the clockwise direction with respect to the body-fixed X-axis, their values must be negative.

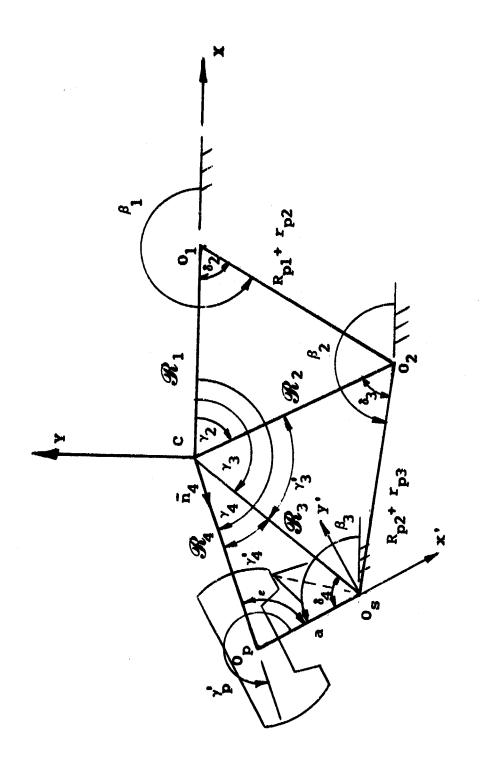


Figure B-4. Fuze body configuration no. 2

Then,

$$\gamma_2 = -\cos^{-1}\left[\frac{R_1^2 + R_2^2 - (R_{p1} + r_{p2})^2}{2R_1R_2}\right]$$
 (B18)

Angle γ_3^* is given by equation B2 and with that the following expression is obtained for γ_3^*

$$\gamma_3 = \gamma_2 - \gamma_3' \tag{B19}$$

The negative sign for γ_3^t is necessary in order to make γ_3 negative.

The angle γ_4^* is given by equation B4. It allows the following expression for angle γ_4^* :

$$\gamma_4 = \gamma_3 - \gamma_4^1 \tag{B20}$$

Again, γ_4^* must have a negative sign to make γ_4 negative.

Angles δ_1 . The angles δ_2 , δ_3 , and δ_4 can be taken directly from equations B6, B7, and B8, respectively.

Angles
$$\beta_1$$
. Angle β_1 becomes:

$$\beta_1 = \pi + \delta_2$$
(B21)

Further

$$\beta_2 = \gamma_2 + \pi + \delta_3 \tag{B22}$$

Finally,

$$\beta_3 = \gamma_3 + \pi + \delta_4 \tag{B23}$$

The angle γ' between the positive x'-axis and the unit vector \vec{n}_4 is shown in figure B-4. This angle is given by:

$$\gamma_p^* = \pi + \varepsilon$$
 (B24)

where

$$\varepsilon = \pi - \delta_{\Delta} - \gamma_{\Delta}^{*} \tag{B25}$$

Therefore

$$\gamma_{p}^{i} = 2\pi - \delta_{4} - \gamma_{4}^{i}$$
 (B26)

The unit vector \vec{n}_4 can again be expressed in terms of the primed coordinate system as

$$\bar{n}_4 = \cos \gamma_p^* \bar{i}^* + \sin \gamma_p^* \bar{j}^*$$
(B27)

The unit vectors $\overline{\mathbf{i}}'$ and $\overline{\mathbf{j}}'$ are expressed in the X-Y system as follows:

$$\vec{1}' = -\cos \beta_3 \vec{1} - \sin \beta_3 \vec{j} \tag{B28}$$

$$\vec{j}' = \sin \beta_3 \vec{i} - \cos \beta_3 \vec{j}$$
 (B29)

Common Computational Expressions for Both Configurations

To find common programming expressions for all angles of both configurations (with the exception of the angles γ^* of equations B14 and B26) the following signum function s_6 is introduced:

This leads to the following expressions:

According to equations Bl and B18:

$$\gamma_2 = s_6 \cos^{-1} \left[\frac{R_1^2 + R_2^2 - (R_{p1} + r_{p2})^2}{2R_1 R_2} \right]$$
 (B32)

According to equation B2:

$$\gamma_3' = \cos^{-1} \left[\frac{R_2^2 + R_3^2 - (R_{p2} + r_{p3})^2}{2R_2R_3} \right]$$
 (B33)

According to equations B3 and B19:

$$\gamma_3 = \gamma_2 + s_6 \gamma_3'$$
 (B34)

According to equation B4:

$$\gamma_{4}^{1} = \cos^{-1} \left[\frac{R_{3}^{2} + R_{4}^{2} - a^{2}}{2R_{3}R_{4}} \right]$$
 (B35)

According to equations B5 and B20:

$$Y_4 = Y_3 + s_6 Y_4$$
 (B36)

According to equations B6, B7, and B8, respectively

$$\delta_2 = \cos^{-1} \left[\frac{(R_{p1} + r_{p2})^2 + R_1^2 - R_2^2}{2R_1(R_{p1} + r_{p2})} \right]$$
 (B37)

$$\delta_3 = \cos^{-1} \left[\frac{(R_{p2} + r_{p3})^2 + 6^2 - 6^2}{2R_2 (R_{p2} + r_{p3})} \right]$$
 (B38)

$$\delta_4 = \cos^{-1} \left[\frac{a^2 + R_3^2 - R_4^2}{2aR_3} \right]$$
 (B39)

Dynamics of Pallet and Escape Wheel in Coupled Motion

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This section shows derivations for both entrance— and exit-coupled motion differential equations. While the resulting combined differential equations are similar, they differ in a number of terms because of the inherent differences of the $\bar{n}_t - \bar{n}_n$ coordinate systems for the entrance and exit sides. While these derivations are given for configuration no. 1, they are also applicable to configuration no. 2 with the help of the signum function s_6 introduced in the previous section.

The analysis of the pallet is performed in the x'-y' system for convenience, while that of the escape wheel uses the X-Y system. An appropriate coordinate transformation makes it possible to combine the systems.

The coefficient of friction μ_1 is used at the escape wheel-pallet interface as well as at the pallet pivot, while the coefficient of friction μ serves to express the friction forces at the escape wheel pivot as well as at the escape wheel-pinion no. 2 tooth interface. (μ is also used in all other gear meshes and gear pivots.)

Signum Functions and Acceleration of Center of Mass of Pallet

To define the direction of the friction force of the escape wheel point S on the coincident pallet point T, the signum function \mathbf{s}_{4} is introduced. Thus,

$$\mathbf{s_4} = \frac{\mathbf{v_{S/T}}}{|\mathbf{v_{S/T}}|} \tag{B40}$$

where $V_{S/T}$ is given by equation C25 of appendix C. The signum function s_5 serves for the determination of the direction of the friction forces acting on the pallet pivot. Then

$$\bullet_5 = \frac{\psi}{|\mathring{\psi}|} \tag{B41}$$

With the constant spin ω of the fuse body, the acceleration \overline{A}_{cp} of the center of mass of the pallet is given by (figs. B-1 and B-5):

$$\vec{A}_{cp} = -\omega^2 R_4 \vec{n}_4 - (\omega + \dot{\psi})^2 r_{cp} [\cos(\psi + \psi_c) \vec{i}' + \sin(\psi + \psi_c) \vec{j}']
+ \ddot{\psi} r_{cp} [-\sin(\psi + \psi_c) \vec{i}' + \cos(\psi + \psi_c) \vec{j}']$$
(B42)

Dynamics of Pallet and Escape Wheel for Entrance-Coupled Motion

A detailed free body diagram of the pallet for entrance-coupled motion is shown in figure B-5. The free body diagram of the escape wheel is simplified and presented for orientation only. A complete free body diagram of the escape wheel which includes all pivot and contact forces is given by figure B-6.

Force Equation of Pallet. Substitution of all forces into Newton's Law, together with the acceleration of the center of mass according to equation B42 furnishes:

$$\begin{split} P_{n}\bar{n}_{n} + \mu_{1}s_{4}P_{n}\bar{n}_{t} + F_{xp}^{*}\bar{1}^{*} - \mu_{1}s_{5}F_{yp}^{*}\bar{1}^{*} + F_{yp}^{*}\bar{j}^{*} + \mu_{1}s_{5}F_{xp}^{*}\bar{j}^{*} \\ &= m_{p} \left\{ -\omega^{2}R_{4}\bar{n}_{4} - (\omega + \dot{\psi})^{2}r_{cp} \left[\cos(\psi + \psi_{c})\bar{1}^{*} + \sin(\psi + \psi_{c})\bar{j}^{*} \right] \right\} \\ &+ \ddot{\psi} r_{cp} \left[-\sin(\psi + \psi_{c})\bar{1}^{*} + \cos(\psi + \psi_{c})\bar{j}^{*} \right] \end{split} \tag{B43}$$

(For nomenclature, consult figures B-1 and B-5, as well as reference B-1.)

Moment Equation of Pallet. The moment equation of the pallet must be written with the understanding that pivot point $\mathbf{0}_{p}$ is an accelerated point. Thus,

$$\vec{\mathbf{M}}_{op} = -\vec{\mathbf{r}}_{op} \times \mathbf{m}_{p} \vec{\mathbf{r}}_{cp} + \dot{\mathbf{H}}_{op}$$
(B44)

where

 \overline{M}_{op} = sum of all external moments with respect to pivot O_p

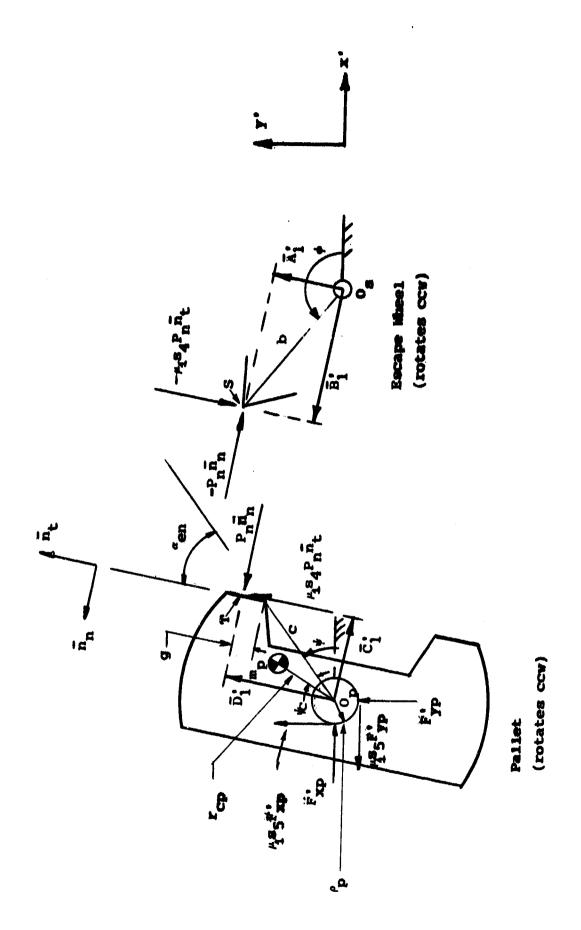


Figure B-5. Free body diagram of pallet in entrance-coupled motion

r_{op} = absolute acceleration of point O_p

 $\dot{H}_{\rm op}$ = time rate of change of angular momentum of the pallet with respect to point $0_{\rm p}$.

Since the spin angular velocity $\bar{\omega}$ of the fuze body is constant, equation B44 takes the following form:

$$\vec{M}_{op} = -(-\omega^2 R_4 \vec{n}_4) \times m_p r_{op} [\cos(\psi + \psi_c) \vec{i}' + \sin(\psi + \psi_c) \vec{j}'] + \vec{I}_p \psi \vec{k}$$
 (B45)

If \overline{n}_4 is substituted according to equation B27, the above becomes in scalar terms:

$$M_{op} = I_{p} \psi - m_{p} r_{cp} R_{4} \omega^{2} \sin(\gamma_{p} - \psi - \psi_{c})$$
 (B46)

To find the value of $\overline{M}_{\mathrm{op}}$ consider figure B-5.

$$\vec{H}_{op} = \vec{D_1} \vec{n_t} \times \vec{P_n} \vec{n_n} + \vec{C_1} \vec{n_n} \times \mu_1 s_4 \vec{P_n} \vec{n_t} - \rho_p \mu_1 s_5 (\vec{F}_{xp} + \vec{F}_{yp}) \vec{k}$$
 (B47)

or, in scalar terms

$$M_{op} = D_1^{\dagger} P_n - C_1^{\dagger} \mu_1 s_4 P_n - \rho_1 \mu_1 s_5 (F_{xp} + F_{yp})$$
 (B48)

When equation B48 is substituted into equation B46, the final moment expression for the pallet is obtained:

$$P_{n} [D_{1}^{i} - C_{1}^{i}\mu_{1}s_{4}] - \rho_{p}\mu_{1}s_{5}(\tilde{F}_{xp} + \tilde{F}_{yp})$$

$$= I_{p}\psi - m_{p}\theta_{4}r_{cp}\omega^{2}sin(\gamma_{p}^{i} - \psi - \psi_{c})$$
(B49)

As originally discussed in an ARRADCOM report (ref B-3), $F_{\rm xp}$ and $F_{\rm yp}$ represent conservatively evaluated pivot force components which assure that the pivot friction moments are opposed to the rotation at all times.

The pivot friction components F'_{xp} and F'_{yp} are first obtained from the following component expressions of equation B43. Subsequently, they are transformed to become F_{xp} and F_{yp} , respectively.

$$-P_{n}\sin(\psi + \alpha) + \mu_{1}s_{4}P_{n}\cos(\psi + \alpha) + F'_{xp} - \mu_{1}s_{5}F'_{yp}$$

$$= m_{p}[-\omega^{2}R_{4}\cos\gamma_{p}' - (\omega + \psi)^{2}r_{cp}\cos(\psi + \psi_{c}) - \psi r_{cp}\sin(\psi + \psi_{c})]$$
(B50)

and

$$P_{n}\cos(\psi + \alpha) + \mu_{1}s_{4}P_{n}\sin(\psi + \alpha) + \mu_{1}s_{5}F_{xp}' + F_{yp}'$$

$$= m_{p} \left[-\omega^{2}R_{4}\sin\gamma_{p}' - (\omega + \psi)^{2}r_{cp}\sin(\psi + \psi_{c}) + \psi r_{cp}\cos(\psi + \psi_{c})\right]$$
(B51)

Simultaneous solution of the above furnishes:

$$\tilde{F}_{yp} = P_n A_1 + \omega^2 A_2 \pm 2 \frac{\omega^2}{|\omega|} \dot{\psi} A_3 \pm \dot{\psi}^2 A_3 \pm \ddot{\psi} A_4$$
 (B52)

$$\tilde{F}_{xp} = P_n A_5 + \omega^2 A_6 \pm 2 \frac{\omega^2}{|\omega|} \dot{\psi} A_7 \pm \dot{\psi}^2 A_7 \pm \dot{\psi} A_8$$
 (B53)

where

$$A_{1} = \left| \frac{-\mu_{1}(s_{4} + s_{5})\sin(\psi + \alpha) - (1 - s_{4}s_{5}\mu_{1}^{2})\cos(\psi + \alpha)}{1 + \mu_{1}^{2}} \right|$$
 (B54)

$$A_{2} = \left| \frac{m_{p} \left[\Re_{4} (\sin \gamma_{p}' - \mu_{1} s_{2} \cos \gamma_{p}') + r_{cp} (\sin (\psi + \psi_{c}) - \mu_{1} s_{2} \cos (\psi + \psi_{c})) \right]}{1 + \mu_{1}^{2}} \right|$$
 (B55)

$$A_3 = \left| \frac{m_p r_{cp} [sin(\psi + \psi_c) - \mu_1 s_5 cos(\psi + \psi_c)]}{1 + \mu_1^2} \right|$$
 (B56)

$$A_4 = \left| \frac{\frac{m_p r_{cp} [\cos(\psi + \psi_c) + \mu_1 s_5 \sin(\psi + \psi_c)]}{1 + \mu_1^2} \right|$$
 (B57)

$$A_5 = \left| \frac{\mu_1(s_4 + s_5)\cos(\psi + \alpha) + (1 - s_4 s_5 \mu_1^2)\sin(\psi + \alpha)}{1 + \mu_1^2} \right|$$
 (B58)

$$A_{6} = \left| \frac{\frac{m_{p} \left[-R_{4} (\cos \gamma_{p}^{*} + \mu_{1} s_{5} \sin \gamma_{p}^{*}) - r_{cp} (\cos (\psi + \psi_{c}) + \mu_{1} s_{5} \sin (\psi + \psi_{c})) \right]}{1 + \mu_{1}^{2}} \right|$$
 (B59)

$$A_7 = \left| \frac{m_p r_{cp} [\cos(\psi + \psi_c) + \mu_1 s_5 \sin(\psi + \psi_c)]}{1 + \mu_1^2} \right|$$
 (B60)

$$A_8 = \left[\frac{m_p r_{cp} [\sin(\psi + \psi_c) - \mu_1 s_5 \cos(\psi + \psi_c)]}{1 + \mu_1^2} \right]$$
 (B61)

The factor $\omega^2/|\omega|$ is introduced in place of ω to assure that the quantity is positive regardless of direction of spin. (This too, assures that the friction moments oppose rotation. The driving moment is proportional to ω and, therefore, independent of spin direction.) To make the final decision concerning the signs in equations B52 and B53, these forces are now substituted into moment equation B49, and the influence of the direction of rotation on each of the resulting moment computations is explored:

$$P_{n}[D_{1}^{!} - C_{1}^{!}\mu_{1}s_{4} - \rho_{p}\mu_{1}s_{5}(A_{1} + A_{5})] \pm \rho_{p}\mu_{1}s_{5}\omega^{2}(A_{2} + A_{6})$$

$$\pm 2\rho_{p}\mu_{1}s_{5}[\omega]^{\psi}(A_{3} + A_{7}) \pm \rho_{p}\mu_{1}s_{5}^{\psi^{2}}(A_{3} + A_{7}) \pm \rho_{p}\mu_{1}s_{5}^{\psi}(A_{4} + A_{8})$$

$$= I_{p}\psi - m_{p}r_{cp}R_{5}\omega^{2}\sin(\gamma_{p}^{*} - \psi - \psi_{c})$$
(B62)

With \mathbf{s}_5 positive for positive rotation, and vice versa, and with all other parameters positive at all times, the following moment components of equation B62 must have negative signs during positive rotation:

$$-P_{n}^{\rho}_{p}^{\mu}_{1}^{s}_{5}(A_{1} + A_{5}) \tag{B63}$$

$$-\rho_{p}\mu_{1}s_{5}\omega^{2}(A_{2}+A_{6})$$
 (B64)

$$-\rho_{\nu}^{\mu_{1}} s_{5}^{\dot{\psi}^{2}} (A_{3} + A_{7}) \tag{A65}$$

The sign of the term containing $\mathring{\psi}$ is negative and is controlled by the sign of $\mathring{\psi}$. Therefore, the signum operator s_5 is omitted and the term becomes:

$$-2\rho_{p}\mu_{1}\frac{\omega^{2}}{|\omega|}(A_{3}+A_{7}) \tag{B66}$$

The choice of sign for the friction moment term in equation B62, which is proportional to the pallet angular acceleration ψ , is discussed in detail in appendix F of reference B-1. This work leads to the computational rules of equations B73 and B74 below which deal with the sign in the expressions for the effective moment of inertia I_{pR} . (The signum function s_5 has now been omitted.)

With these considerations, equation $B62^{B-1}$ now becomes

$$P_{n}^{A}_{18} - \omega^{2}_{19} - 2\frac{\omega^{2}}{|\omega|} \dot{\psi}_{20} - \dot{\psi}^{2}_{A_{21}}$$

$$= I_{pR} \dot{\psi} - m_{p} r_{cp}^{R}_{4} \omega^{2} \sin(\gamma_{p}^{*} - \psi - \psi_{c})$$
(B68)

where B-2

$$A_{18} = D_1^t - C_1^t \mu_1 s_4 - \rho_D \mu_1 s_5 (A_1 + A_5)$$
 (B69)

$$A_{19} = \rho_{p} \mu_{1} s_{5} (A_{2} + A_{6})$$
 (B70)

$$A_{20} = \rho_{p} \mu_{1} (A_{3} + A_{7}) \tag{B71}$$

$$A_{21} = \rho_{p} \mu_{1} s_{5} (A_{3} + A_{7}) \tag{B72}$$

$$I_{PR} = I_p + A_{22}$$
, when ψ and ψ have identical signs (B73)

$$I_{PR} = I_p - A_{22}$$
, when $\dot{\psi}$ and $\dot{\psi}$ have opposite signs^{B-3} (B74)

$$A_{22} = \rho_{p} \mu_{1} (A_{4} + A_{8}) \tag{B75}$$

Equation B68 is now used to find an expression for the contact force \mathbf{P}_n . This expression will later make it possible to establish a single differential equation for the escapement in coupled motion. Thus,

$$P_{n} = \frac{I_{PR}^{"} + A_{21}^{"} + 2 \frac{\omega^{2}}{|\omega|} A_{20}^{"} + \omega^{2} A_{19} - m_{p} r_{cp}^{R} \omega^{2} \sin(\gamma_{p}^{"} - \psi - \psi_{c})}{A_{18}}$$
(B76)

 $^{^{\}mathrm{B-1}}$ Note that there is no equation B67.

 $^{^{}B-2}$ The parameters A are not sequential at this point. A 49 to A 17 may be found in equations B 890 to B 595 and B 597 to B 599 in conjunction with the work on the escape wheel.

 $^{^{}B-3}$ Care must be taken that $\rm I_p$ - $\rm A_{22}$ does not become negative. If this occurs, $\rm I_{PR}$ must be set equal to zero.

It is now necessary to write equation B76 in terms of the angular velocity $\dot{\phi}$ and the angular acceleration ϕ of the escape wheel so that it may later be equated to a similar expression for the escape wheel.

Thus, with the help of equations C19 and C26 of appendix C, the contact force becomes:

$$P_{n} = \frac{1}{A_{18}} \left[I_{PR} U \dot{\phi} + (A_{21} U^{2} + I_{PR} V) \dot{\phi}^{2} + 2 \frac{\omega^{2}}{|\omega|} A_{20} U \dot{\phi} + \omega^{2} A_{19} - m_{p} r_{cp} \alpha_{4} \omega^{2} \sin(\gamma_{p}^{*} - \psi - \psi_{c}) \right]$$
(B77)

Escape Wheel Analysis. A detailed free body diagram of the escape wheel and pinion no. 3 in entrance-coupled motion is shown in figure B-6. The pivot forces $\mathbf{F}_{\mathbf{x}3}$ and $\mathbf{F}_{\mathbf{y}3}$ as well as the contact force \mathbf{F}_{23} and the force \mathbf{T}_3 are now defined in the general X-Y system. This makes it necessary to transform the unit vectors \mathbf{n}_t and \mathbf{n}_n from the x' - y' system to the X-Y system. Since

$$\vec{1}' = \cos(\beta_3 - \pi)\vec{1} + \sin(\beta_3 - \pi)\vec{j}$$
 (B78)

or

$$\vec{i}' = -\cos\beta_3 \vec{i} - \sin\beta_3 \vec{j}, \tag{B79}$$

and

$$\vec{j}' = \sin\beta_3 \vec{i} - \cos\beta_3 \vec{j}$$
 (B80)

the unit vectors \overline{n}_t and \overline{n}_n can now be determined with the help of equations C5 and C6 of appendix C. Thus,

$$\bar{n}_{t} = -\cos(\psi + \alpha + \beta_{3})\bar{i} - \sin(\psi + \alpha + \beta_{3})\bar{j}$$
(B81)

$$\vec{n}_n = \sin(\psi + \alpha + \beta_3)\vec{1} - \cos(\psi + \alpha + \beta_3)\vec{j}$$
 (B82)

Parallel work is given in appendix B of reference B-1.

Force Equation of Escape Wheel. Force equilibrium is obtained with the help of the following expression, based on figure B-6:

$$-P_{n}\bar{n}_{n} - \mu_{1} *_{4} P_{n}\bar{n}_{c} + F_{23}\bar{n}_{23} + \mu *_{2} F_{23}\bar{n}_{N23} + \bar{T}_{3}$$

$$+ F_{x3}\bar{i} + \mu F_{y3}\bar{i} + \mu F_{x3}\bar{j} - F_{y3}\bar{j} = 0$$
(B83a)

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Figure B-6. Free body diagram of escape wheel and pinion 3 in entrance-coupled motion

The signum function s_2 is defined in in reference B-3. Further, according to reference B-1, the D'Alembert force T_3 is given by

$$\overline{T}_3 = m_3 R_3 \omega^2 (\cos \gamma_3 \overline{1} + \sin \gamma_3 \overline{j})$$
 (B83b)

$$A_{1}^{'}\bar{n}_{t}x(-P_{n})\bar{n}_{n} + B_{1}^{'}\bar{n}_{n} \times (-\mu_{1}s_{4}P_{n})\bar{n}_{t}$$

$$-\mu\rho_{3}(\tilde{F}_{x3} + \tilde{F}_{y3})\bar{k} + r_{b3}F_{23}\bar{k} - \mu s_{2}(d_{2} - a_{2})F_{23}\bar{k} = I_{3}\bar{\phi}\bar{k}$$
(B84)

This becomes in scalar form:

$$-P_{n}(A_{1}' - B_{1}'\mu_{1}B_{4}) - \mu\rho_{3}(\tilde{F}_{x3} + \tilde{F}_{y3}) + r_{b3}F_{23}$$

$$-\mu B_{2}(d_{2} - B_{2})F_{23} = I_{3}\ddot{\phi}$$
(B85)

The pivot force components F_{x3} and F_{y3} , which eventually become F_{x3} and F_{y3} , are now determined with the help of the following scalar component expressions of equation B83:

$$-P_{n}\sin(\psi + \alpha + \beta_{3}) + \mu_{1}s_{4}P_{n}\cos(\psi + \alpha + \beta_{3}) + T_{3}\cos\gamma_{3}$$

$$+ F_{23}\sin(\beta_{2} + \theta_{2}) + \mu s_{2}F_{23}\cos(\beta_{2} + \theta_{2}) + F_{x3} + \mu F_{y3} = 0$$
(B86)

and

$$P_{n}\cos(\psi + \alpha + \beta_{3}) + \mu_{1}s_{4}P_{n}\sin(\psi + \alpha + \beta_{3}) + T_{3}\sin\gamma_{3}$$

$$-F_{23}\cos(\beta_{2} + \theta_{2}) + \mu_{2}F_{23}\sin(\beta_{2} + \theta_{2}) + \mu_{x3} - F_{y3} = 0$$
(B87)

Simultaneous solution of the above yields the following "conservative" pivot force components:

$$\tilde{F}_{y3} = P_n A_9 + F_{23} A_{10} + T_3 A_{11}$$
 (B88)

$$\tilde{F}_{x3} = P_n A_{12} + F_{23} A_{13} + T_3 A_{14}$$
 (B89)

Since there are no kinematic terms, it is most conservative to sum all terms and use the absolute values of ${\rm A_9-A_{14}}$ as follows:

$$A_9 = \left| \frac{(\mu_1 s_4 + \mu) \sin(\psi + \alpha + \beta_3) + (1 - \mu \mu_1 s_4) \cos(\psi + \alpha + \beta_3)}{1 + \mu^2} \right|$$
 (B90)

$$A_{10} = \left| \frac{-\mu(1-s_2)\sin(\beta_2 + \theta_2) - (1 + \mu^2 s_2)\cos(\beta_2 + \theta_2)}{1 + \mu^2} \right|$$
 (B91)

$$A_{11} = \left| \frac{\sin \gamma_3 - \mu \cos \gamma_3}{1 + \mu^2} \right|$$
 (B92)

$$A_{12} = \left| \frac{(1 - \mu_1 \mu s_4) \sin(\psi + \alpha + \beta_3) - (s_4 \mu_1 + \mu) \cos(\psi + \alpha + \beta_3)}{1 + \mu^2} \right|$$
(B93)

$$A_{13} = \left| \frac{-(1 + \mu^2 s_2) \sin(\beta_2 + \theta_2) + \mu(1 - s_2) \cos(\beta_2 + \theta_2)}{1 + \mu^2} \right|$$
 (B94)

$$A_{14} = \left| \frac{\cos \gamma_3 + \mu \sin \gamma_3}{1 + \mu^2} \right|$$
 (B95)

Substitution of equations B88 and B89 into the moment equation B85 and subsequent rearrangement lead to the following expression for the contact force P_n in terms of escape wheel quantities:

$$P_{n} = \frac{-I_{3}^{"} + F_{23}^{A}_{15} - T_{3}^{A}_{16}}{A_{17}}$$
 (B96)

where

$$A_{15} = r_{b3} - \mu[s_2(d_2 - a_2) + \rho_3(A_{10} + A_{13})]$$
 (B97)

$$A_{16} = \mu \rho_3 (A_{11} + A_{14})$$
 (B98)

$$A_{17} = A_1' - \mu_1 s_4 B_1' + \mu \rho_3 (A_9 + A_{12})$$
 (B99)

Combined Entranced-Coupled Motion Differential Equation. Equations B77 and B96 are now set equal to each other. This furnishes the following combined motion differential equation for the escapement:

$$[I_{3}A_{18} + I_{PR}A_{17}U]\ddot{\phi} + [I_{PR}A_{17}V + A_{17}A_{21}U^{2}]\dot{\phi}^{2} + 2\frac{\omega^{2}}{|\omega|}A_{17}A_{20}U\dot{\phi}$$

$$= A_{15}A_{18}F_{23} - A_{16}A_{18}T_{3} - A_{17}\omega^{2}[A_{19} - m_{p}r_{cp}A_{4}\sin(\gamma_{p}^{*} - \psi - \psi_{c})]$$
(B100)

Dynamics of Pallet and Escape Wheel for Exit-Coupled Motion

A free body diagram of the pallet for exit-coupled motion, together with a simplified representation of the escape wheel is shown in figure B-7. When this figure is compared with figure B-5, which depicts entrance-coupled motion, it is noted that now

$$\vec{P}_n = -P_n \vec{n}_n \tag{B101}$$

This sign change is reflected both in the force and in the moment equation.

Force Equations of Pallet. When the scalar force equations B50 and B51 are altered to account for the above change, they have the following form:

$$P_{n}\sin(\psi + \alpha) + \mu_{1}s_{4}P_{n}\cos(\psi + \alpha) + F'_{xp} - \mu_{1}s_{5}F'_{yp}$$

$$= m_{p}[-\omega^{2}R_{4}\cos\gamma_{p}' - (\omega + \dot{\psi})^{2}r_{cp}\cos(\psi + \psi_{c}) - \dot{\psi}r_{cp}\sin(\psi + \psi_{c})]$$
(B102)

$$-P_{n}\cos(\psi + \alpha) + \mu_{1}s_{4}P_{n}\sin(\psi + \alpha) + \mu_{1}s_{5}F_{xp}' + F_{yp}'$$

$$= m_{p}[-\omega^{2}R_{4}\sin\gamma_{p}' - (\omega + \dot{\psi})^{2}r_{cp}\sin(\psi + \psi_{c}) + \psi r_{cp}\cos(\psi + \psi_{c})]$$
(B103)

Simultaneous solution of the above shows that only the factors $\rm A_1$ and $\rm A_5$ as given by equations B54 and B58 need be changed. These will now be called AA $_1$ and AA $_5$. Then,

$$AA_{1} = \left| \frac{\mu_{1}(s_{5} - s_{4})\sin(\psi + \alpha) + (1 + \mu_{1}^{2}s_{4}s_{5})\cos(\psi + \alpha)}{1 + \mu_{1}^{2}} \right|$$
 (B104)

and

$$AA_5 = \left| \frac{\mu_1(s_5 - s_4)\cos(\psi + \alpha) - (1 + \mu_1^2 s_4 s_5)\sin(\psi + \alpha)}{1 + \mu_1^2} \right|$$
 (B105)

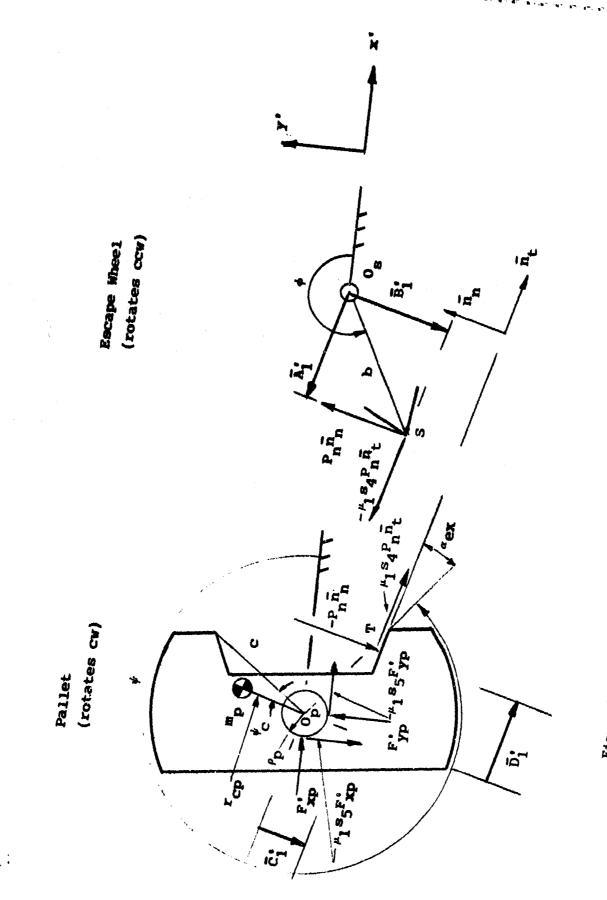


Figure B-7, Free body diagram of pallet in exit-coupled motion

$$P_{n}[-D_{1}^{\dagger} - C_{1}^{\dagger}\mu_{1}s_{4}] - \rho_{p}\mu_{1}s_{5}(\tilde{F}_{xp} + \tilde{F}_{yp})$$

$$= I_{p}\tilde{\psi} - m_{p}R_{4}r_{cp}\omega^{2}sin(\gamma_{p}^{\dagger} - \psi - \psi_{c})$$
(B106)

(Note change of sign of D;.)

Following the same path as for the entrance contact, it can be shown that, with the exception of the factor A_{18} , equation B76 for the contact force P_n remains unchanged. Then,

$$P_{n} = \frac{I_{pR} \ddot{\psi} + A_{21} \dot{\psi}^{2} + 2 \frac{\omega^{2}}{|\omega|^{A} 20} \dot{\psi} + \omega^{2} A_{19} - m_{p} r_{cp} R_{4} \omega^{2} \sin(\gamma_{p}^{*} - \psi - \psi_{c})}{AA_{18}}$$
(B107)

where

$$AA_{18} = -[D_1^t + C_1^t \mu_1 s_4 + \rho_p \mu_1 s_5 (AA_1 + AA_5)]$$
 (B108)

Finally, the expression equivalent to equation B77 becomes:

$$P_{n} = \frac{1}{AA_{18}} \left[I_{PR} U_{\phi}^{"} + (A_{21} U^{2} + I_{PR} V)_{\phi}^{"}^{2} + 2 \frac{\omega^{2}}{|\omega|} A_{20} U_{\phi}^{"} + \omega^{2} A_{19} - m_{p} r_{cp} R_{4} \omega^{2} \sin(\gamma_{p}^{"} - \psi - \psi_{c}) \right]$$
(B109)

Force Equations of Escape Wheel. A free body diagram of the escape wheel in exit-coupled motion is shown in figure B-8. When the scalar force equations B86 and B87 for the escape wheel in entrance-coupled motion are adjusted to account for this condition, the following expressions are obtained:

$$P_{n}\sin(\psi + \alpha + \beta_{3}) + \mu_{1}s_{4}P_{n}\cos(\psi + \alpha + \beta_{3}) + T_{3}\cos\gamma_{3} + F_{23}\sin(\beta_{2} + \theta_{2}) + \mu s_{2}F_{23}\cos(\beta_{2} + \theta_{2}) + F_{x3} + \mu F_{y3} = 0$$
 (B110)

and

$$-P_{n}\cos(\psi + \alpha + \beta_{3}) + \mu_{1}s_{4}P_{n}\sin(\psi + \alpha + \beta_{3}) + T_{3}\sin\gamma_{3}$$

$$-F_{23}\cos(\beta_{2} + \theta_{2}) + \mu s_{2}F_{23}\sin(\beta_{2} + \theta_{2}) + \mu F_{x3} - F_{y3} = 0$$
 (B111)

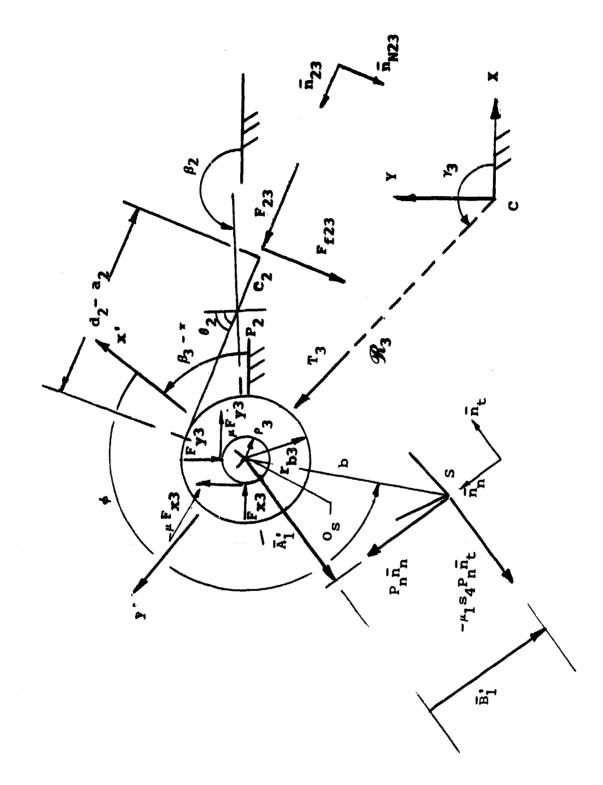


Figure B-8. Free body diagram of escape wheel and pinion 3 in exit-coupled motion

Simultaneous solution of the above for F_{x3} and F_{y3} shows that certain coefficients of these pivot forces will differ from those given in the solutions for entrance-coupled motion. Thus,

$$F_{y3} = P_n AA_9 + F_{23}A_{10} + T_3A_{11}$$
 (B112)

$$\tilde{F}_{x3} = P_n AA_{12} + F_{23}A_{13} + T_3A_{14}$$
 (B113)

where A_{10} , A_{11} , A_{13} , and A_{14} are given by equations B91, B92, B94, and B95, respectively. The new coefficients AA_9 and AA_{12} are defined as follows:

$$AA_9 = \left| \frac{(\mu_1 s_4 - \mu) \sin(\psi + \alpha + \beta_3) - (1 + s_4 \mu \mu_1) \cos(\psi + \alpha + \beta_3)}{1 + \mu^2} \right|$$
(B114)

$$AA_{12} = \left| \frac{-(1 + \mu \mu_1 s_4) \sin(\psi + \alpha + \beta_3) + (\mu - \mu_1 s_4) \cos(\psi + \alpha + \beta_3)}{1 + \mu^2} \right|$$
 (B115)

Moment Equation of Escape Wheel. Modification of the escape wheel moment equation B84 for exit contact gives:

$$A_{1}^{\prime} \vec{n}_{t} \times P_{n} \vec{n}_{n} + \vec{B}_{1}^{\prime} \times (-\mu_{1} s_{4} P_{n}) \vec{n}_{t} - \mu \rho_{3} (\vec{F}_{x3} + \vec{F}_{y3}) \vec{k}$$

$$+ r_{h3} F_{23} \vec{k} - \mu s_{2} (d_{2} - a_{2}) F_{23} \vec{k} = I_{3} \phi \vec{k}$$
(B116)

Substitution of equations B112 and B113 into the scalar form of equation B116 above furnishes the following expression for $P_{\rm m}$:

$$P_{n} = \frac{-I_{3}^{\phi} + F_{23}^{A}_{15} - I_{3}^{A}_{16}}{AA_{17}}$$
 (B117)

where A_{15} and A_{16} are given by equations B97 and B98, respectively. The factor AA_{17} is new; i.e.,

$$AA_{17} = -A_1' - \mu_1 s_4 B_1' + \mu \rho_3 (AA_9 + AA_{12})$$
 (B118)

Combined Exit-Coupled Motion Differential Equation. Equations B109 and B117 are now equated to each other. This furnishes the combined exit-coupled motion differential equation:

$$[I_{3}AA_{18} + I_{pR}AA_{17}U]^{\circ} + [I_{pR}AA_{17}V + AA_{17}A_{21}U^{2}]^{\circ}^{2}$$

$$+ 2 \frac{\omega^{2}}{|\omega|} AA_{17}A_{20}U^{\circ}$$

$$- A_{15}AA_{18}F_{23} - A_{16}AA_{18}T_{3} - AA_{17}\omega^{2}[A_{19} - m_{p}r_{cp}G_{4}sin(\gamma_{p}^{*} - \psi - \psi_{c})]$$
(B119)

Dynamics of Rotor (Gear No. 1) (Applicable to both configurations with appropriate choice of s₆. See also reference B-1.)

A free body diagram of the rotor of configuration no. 1, which moves in a counterclockwise direction, is shown in figure B-9. The acceleration of its center of mass is given by equation Al22 of appendix A of reference B-1.

Since the motion of the rotor must be expressed in terms of the escape wheel variable ϕ , use must be made of the following relationships:

$$\hat{\phi}_1 = N_{31}\hat{\phi} \tag{B120}$$

and

where

$$N_{31} = \frac{N_{P3} \times N_{P2}}{N_{C2} \times N_{C1}}$$
 (B122)

The rotor angle ϕ_{1C} + ϕ_{1} is expressed in terms of the total escape wheel rotation ϕ_{T} :

$$\phi_{1C} + \phi_1 = \phi_{1C} + N_{31} \phi_T$$
 (B123)

■ 大学はは、1911年 | 株式の名の名を乗り出ているのである。 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 | 1911年 |

(For details, see Program Description.)

Newton's force equation may now be written in the form of equation B63 of reference B-1. The tooth forces

$$\vec{F}_{21} = -F_{12} \vec{n}_{12}$$
 (B124)

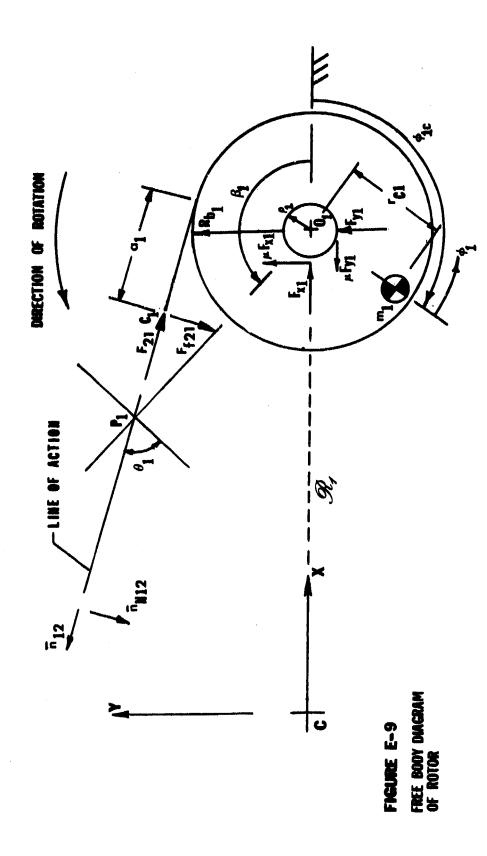


Figure B-9. Free body diagram of rotor

and

$$\vec{F}_{f21} = \mu s_1 \vec{F}_{12} \vec{n}_{N12}$$
 (B125)

have the appropriate directions to account for friction. (The signum function s_1 is defined in reference B-1.)

Thus:

$$\begin{aligned} & -F_{21}\bar{n}_{12} + \mu s_{1}F_{12}\bar{n}_{N12} + F_{x1}\bar{i} - \mu F_{y1}\bar{i} + F_{y1}\bar{j} + \mu F_{x1}\bar{j} \\ &= m_{1} \{ -\omega^{2}R_{1}\bar{i} - (\omega + N_{31}\dot{\phi})^{2}r_{c1} [\cos(\phi_{1c} + N_{31}\phi_{T})\bar{i} + \sin(\phi_{1c} + N_{31}\phi_{T})\bar{j}] \\ &+ N_{31}\phi r_{c1} [-\sin(\phi_{1c} + N_{31}\phi_{T})\bar{i} + \cos(\phi_{1c} + N_{31}\phi_{T})\bar{j}] \} \end{aligned}$$
(B126)

where

$$\vec{n}_{12} = -\sin(\beta_1 - \theta_1)\vec{i} + \cos(\beta_1 - \theta_1)\vec{j}$$
(B127)

and

$$\vec{n}_{N12} = -\cos(\beta_1 - \theta_1)\vec{i} - \sin(\beta_1 - \theta_1)\vec{j}$$
 (B128)

The moment equation must be written with respect to the accelerated point 0_1 . The pivot reactions $F_{\chi l}$ and $F_{\chi l}$ are written such that the associated friction moments retard the counterclockwise rotation of the rotor. This leads to:

$$-F_{12}R_{b1} + \mu_{B_{1}a_{1}}F_{12} - \mu_{P_{1}}(\tilde{F}_{x1} + \tilde{F}_{y1})$$

$$= m_{1}\omega^{2}R_{1}r_{c1}\sin(\phi_{1c} + N_{31}\phi_{T}) + I_{1}N_{31}\phi$$
(B129)

The forces F_{χ^1} and F_{χ^1} are obtained after the simultaneous solution of the following component expressions of equation B63 of reference B-1 for F_{χ^1} and F_{χ^1} :

$$F_{12}\sin(\beta_{1}-\theta_{1}) - \mu s_{1}F_{12}\cos(\beta_{1}-\theta_{1}) + F_{x1} - \mu F_{y1}$$

$$= m_{1}[-\omega^{2}\theta_{1}^{2} - (\omega + N_{31}\dot{\phi})^{2}r_{c1}\cos(\phi_{1c} + N_{31}\phi_{T}) - N_{31}\phi r_{c1}\sin(\phi_{1c} + N_{31}\phi_{T})]$$
(B130)

and

$$-F_{12}\cos(\beta_{1} - \theta_{1}) - \mu s_{1}F_{12}\sin(\beta_{1} - \theta_{1}) + F_{y1} + \mu F_{x1}$$

$$= m_{1}[-(\omega + N_{31}\dot{\phi})^{2}r_{c1}\sin(\phi_{1c} + N_{31}\phi_{T}) + N_{31}\phi_{rc1}\cos(\phi_{1c} + N_{31}\phi_{T})]$$
(B131)

Simultaneous solution of the above furnishes:

$$\tilde{F}_{x1} = F_{12}^{A}_{23} \pm \omega^{2}_{24} \pm 2\frac{\omega^{2}}{|\omega|} N_{31}^{\phi}_{A_{25}} \pm (N_{31}^{\phi})^{2}_{A_{25}} \pm N_{31}^{\phi}_{A_{26}}$$
(B132)

$$\tilde{F}_{y1} = F_{12}A_{27} \pm \omega^2 A_{28} \pm 2\frac{\omega^2}{|\omega|} N_{31} \phi A_{29} \pm (N_{31} \phi)^2 A_{29} \pm N_{31} \phi A_{30}$$
 (B133)

where

$$A_{23} = \left| \frac{(1 - \mu^2 s_1) \sin(\beta_1 - \theta_1) - \mu(1 + s_1) \cos(\beta_1 - \theta_1)}{1 + \mu^2} \right|$$
 (B134)

$$A_{24} = \left| \frac{m_1[R_1 + r_{c1}(\cos(\phi_{1c} + N_{31}\phi_{T}) + \mu\sin(\phi_{1c} + N_{31}\phi_{T}))]}{1 + \mu^2} \right|$$
 (B135)

$$A_{25} = \left| \frac{m_1 r_{c1} \left[\cos(\phi_{1c} + N_{31} \phi_{T}) + \mu \sin(\phi_{1c} + N_{31} \phi_{T}) \right]}{1 + \mu^2} \right|$$
 (B136)

$$A_{26} = \left| \frac{m_1 r_{c1} \left[\sin(\phi_{1c} + N_{31} \phi_{T}) - \mu \cos(\phi_{1c} + N_{31} \phi_{T}) \right]}{1 + \mu^2} \right|$$
 (B137)

$$A_{27} = \left| \frac{\mu(1+s_1)\sin(\beta_1-\theta_1) + (1-\mu^2s_1)\cos(\beta_1-\theta_1)}{1+\mu^2} \right|$$
 (B138)

$$A_{28} = \left| \frac{\mu m_1 R_1 + m_1 r_{c1} \left[\mu \cos(\phi_{1c} + N_{31} \phi_{T}) - \sin(\phi_{1c} + N_{31} \phi_{T}) \right]}{1 + \mu^2} \right|$$
 (B139)

$$A_{29} = \left| \frac{m_1 r_{c1} \left[\mu \cos(\phi_{1c} + N_{31} \phi_{T}) - \sin(\phi_{1c} + N_{31} \phi_{T}) \right]}{1 + \mu^2} \right|$$
 (B140)

$$A_{30} = \left| \frac{m_1 r_{c1} \left[\mu sin(\phi_{1c} + N_{31}\phi_{T}) + cos(\phi_{1c} + N_{31}\phi_{T}) \right]}{1 + \mu^2} \right|$$
(B141)

Equations B132 and B133 are now substituted into the moment equation B129.

$$-F_{12}R_{b1} + \mu s_{1}a_{1}F_{12} - \mu \rho_{1}[F_{12}(A_{23} + A_{27}) \pm \omega^{2}(A_{24} + A_{28})]$$

$$\pm 2\frac{\omega^{2}}{|\omega|}N_{31}^{3}(A_{25} + A_{29}) \pm (N_{31}^{3})^{2}(A_{25} + A_{29}) \pm N_{31}^{3}(A_{26} + A_{30})]$$

$$= m_{1}\omega^{2}R_{1}r_{c1}sin(\phi_{1c} + N_{31}\phi_{T}) + I_{1}N_{31}^{3}\phi$$
(B142)

In order to have the pivot friction moment negative, the following terms on the left hand side of equation B142 must be negative:

$$\mu\rho_{1} F_{12} (A_{23} + A_{27})$$
 $\mu\rho_{1} \omega^{2} (A_{24} + A_{28})$
 $\mu\rho_{1} (N_{31} \dot{\phi})^{2} (A_{25} + A_{29})$
 $2|\mu|\rho_{1} \frac{\omega^{2}}{|\omega|} N_{31} \dot{\phi} (A_{25} + A_{29})$
 $B-4$

The term $\mu\rho_1N_{31}\phi(A_2+A_{30})$ is treated in the same manner as was shown in connection with equation Al47 of reference B-1. (The discussion in appendix A of reference B-1 is also of interest for the determination of the effective moment of inertia I_{1R} .)

The above considerations give the moment equation the following form:

$$-F_{12}[R_{b1} - \mu s_{1}a_{1} + \mu \rho_{1}(A_{23} + A_{27})] - \omega^{2}\mu \rho_{1}(A_{24} + A_{28})$$

$$-\frac{2\mu \rho_{1}\omega^{2}N_{31}^{\dagger}}{|\omega|}(A_{25} + A_{29}) - \mu \rho_{1}(N_{31}^{\dagger})^{2}(A_{25} + A_{29})$$

$$= m_{1}\omega^{2}R_{1}r_{c1}\sin(\phi_{1c} + N_{31}\phi_{T}) + [I_{1} \pm \mu \rho_{1}(A_{26} + A_{30})]N_{31}^{\dagger}$$
(B143)

 $^{^{}B-4}$ With N₃₁ $\dot{\phi}$ = positive, must be absolute. See also discussion following equation Al47 of reference B-1.

Finally, the above expression is solved for \mathbf{F}_{12} , using the effective moment of inertia \mathbf{I}_{1R} :

$$F_{12} = \frac{-A_{32} - A_{33}N_{31}\dot{\phi} - A_{34}(N_{31}\dot{\phi})^2 - A_{35}\sin(\phi_{1c} + N_{31}\phi_{T}) - I_{1R}N_{31}\dot{\phi}}{A_{31}}$$
(B144)

where

$$A_{31} = R_{b1} - \mu s_{1} a_{1} + \mu \rho_{1} (A_{23} + A_{27})$$
 (B145)

$$A_{32} = \mu \rho_1 \omega^2 (A_{24} + A_{28}) \tag{B146}$$

$$A_{33} = 2|\mu|\rho_1 \frac{\omega^2}{|\omega|} (A_{25} + A_{29})$$
 (B147)

$$A_{34} = \mu \rho_1 (A_{25} + A_{29})$$
 (B148)

$$A_{35} = m_1 R_{1} r_{c1} \omega^2$$
 (B149)

$$I_{1R} = I_1 + |\mu|\rho_1(A_{26} + A_{30})$$
, when ϕ and ϕ have the same signs. (B150)

$$I_{1R} = I_1 - |\mu|\rho_1 (A_{26} + A_{30})$$
, when $\dot{\phi}$ and $\dot{\phi}$ have opposite signs. (B151)

Dynamics of Gear and Pinion Set No. 2 (applicable to both configurations with appropriate choice of \mathbf{s}_6)

The free body diagram of gear and pinion set no. 2 is shown in figure B-10. Its force equation includes the D'Alembert force

$$T_2 = m_2 R_2 \omega^2 \tag{B152}$$

Thus,

$$-F_{23}\bar{n}_{23} - s_{2}\mu F_{23}\bar{n}_{N23} + F_{12}\bar{n}_{12} - \mu s_{1}F_{12}\bar{n}_{N12} + F_{x2}\bar{i} + \mu F_{y2}\bar{i}$$

$$+ F_{y2}\bar{j} - \mu F_{x2}\bar{j} + T_{2}(\cos \gamma_{2}\bar{i} + \sin \gamma_{2}\bar{j}) = 0$$
(B153)

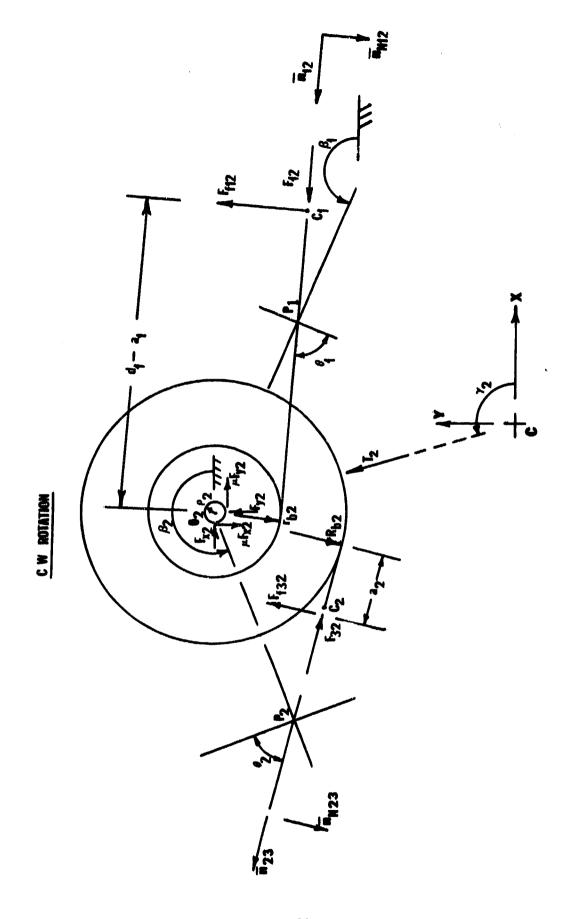


Figure B-10. Free body diagram of gear and pinion set no. 2

where

$$\vec{n}_{23} = \sin(\beta_2 + \theta_2)\vec{i} - \cos(\beta_2 + \theta_2)\vec{j}$$
 (B154)

and

$$\vec{n}_{N23} = \cos(\beta_2 + \theta_2)\vec{i} + \sin(\beta_2 + \theta_2)\vec{j}$$
 (B155)

The unit vectors \overline{n}_{12} and \overline{n}_{N12} were given by equations Bl27 and Bl28. Hence

$$\vec{F}_{32} = -F_{23} \vec{n}_{23}$$
 (B156)

$$\bar{\mathbf{F}}_{\mathbf{f}32} = -\mathbf{s}_2 \mu \bar{\mathbf{F}}_{23} \bar{\mathbf{n}}_{N23}$$
 (B157)

To write the moment equation, let

$$\phi_2 = N_{32}\phi$$
 (B158)

where

$$N_{32} = -\frac{N_{P3}}{N_{G2}}$$
 (B159)

Then,

$$F_{23}^{R}_{b2} = \mu s_{2}^{F}_{23}^{a_{2}} = F_{12}^{F}_{b2} + \mu s_{1}^{F}_{12}(d_{1} - a_{1}) + \mu \rho_{2}(F_{x2} + F_{y2})$$
 (B160)
= $I_{2}^{N}_{32}^{\phi}$

The conservative bearing forces F_{x2} and F_{y2} are obtained from the simultaneous solution of the following component equations of the force expression B153 for F_{x2} and F_{y2} :

$$-F_{23}\sin(\beta_2 + \theta_2) - F_{23}\mu s_2\cos(\beta_2 + \theta_2) - F_{12}\sin(\beta_1 - \theta_1)$$

$$+\mu s_1 F_{12}\cos(\beta_1 - \theta_1) + F_{x2} + \mu F_{y2} + T_2\cos\gamma_2 = 0$$
(B161)

and

$$F_{23}\cos(\beta_2 + \theta_2) - \mu s_2 F_{23}\sin(\beta_2 + \theta_2) + F_{12}\cos(\beta_1 - \theta_1)$$

$$+\mu s_1 F_{12}\sin(\beta_1 - \theta_1) + F_{y2} - \mu F_{y2} + T_2\sin\gamma_2 = 0$$
(B162)

This leads to:

$$F_{x2} = F_{23}A_{36} + F_{12}A_{37} + T_{2}A_{38}$$
 (B163)

$$F_{y2} = F_{23}A_{39} + F_{12}A_{40} + T_{2}A_{41}$$
 (B164)

(All signs are left positive in order to furnish positive friction moments in equation B160.) In the above:

$$A_{36} = \left| \frac{(1 - \mu^2 s_2) \sin(\beta_2 + \theta_2) + \mu(1 + s_2) \cos(\beta_2 + \theta_2)}{1 + \mu^2} \right|$$
 (B165)

$$A_{37} = \left| \frac{(1 + \mu^2 s_1) \sin(\beta_1 - \theta_1) + \mu(1 - s_1) \cos(\beta_1 - \theta_1)}{1 + \mu^2} \right|$$
 (B166)

$$A_{38} = \left| \frac{\cos \gamma_2 - \mu \sin \gamma_2}{1 + \mu^2} \right|$$
 (B167)

$$A_{39} = \left| \frac{\mu(1+s_2)\sin(\beta_2+\theta_2) - (1-\mu^2s_2)\cos(\beta_2+\theta_2)}{1+\mu^2} \right|$$
 (B168)

$$A_{40} = \left| \frac{\mu(1-s_1)\sin(\beta_1-\theta_1) - (1+\mu^2s_1)\cos(\beta_1-\theta_1)}{1+\mu^2} \right|$$
 (B169)

$$A_{41} = \left| \frac{\mu \cos \gamma_2 + \sin \gamma_2}{1 + \mu^2} \right|$$
 (B170)

Substitution of equations B163 and B164 into the moment equation B160 and subsequent solution for the contact force ${\bf F}_{23}$ give:

$$F_{23} = \frac{F_{12}A_{42} - T_{2}A_{43} + I_{2}N_{32}\phi}{A_{44}}$$
(B171)

where

$$A_{42} = r_{b2} - \mu[s_1(d_1 - a_1) + \rho_2(A_{37} + A_{40})]$$
 (B172)

$$A_{43} = \mu \rho_2 (A_{38} + A_{41})$$
 (B173)

$$A_{44} = R_{b2} - \mu[s_2 a_2 - \rho_2(A_{36} + A_{39})]$$
 (B174)

Dynamics of Combined System in Coupled Motion (applicable to both configurations)

To obtain a single differential equation for the total system in coupled motion in terms of the escape wheel angle ϕ , an appropriate expression for the contact force F_{23} , which also contains the contribution of the rotor, must be substituted into either equation B100 or equation B119 for entrance or exit-coupled motion, respectively.

Thus, first substitute equation B144 for F_{12} into equation B171, which is the above expression for F_{23} :

$$F_{23} = \frac{1}{A_{44}} \left[\frac{-A_{42}A_{32}}{A_{31}} - T_{2}A_{43} - \frac{A_{42}A_{33}}{A_{31}} N_{31}^{2} - \frac{A_{42}A_{34}}{A_{31}} N_{31}^{2} \right]^{2}$$

$$- \frac{A_{42}A_{35}}{A_{31}} \sin(\phi_{1c} + N_{31}\phi_{T}) + (I_{2}N_{32} - \frac{A_{42}}{A_{31}} I_{1R}N_{31})^{2}$$
(B175)

Now, equation B175 is substituted into equation B100 or equation B119 and the final differential equation of coupled motion results:

$$A_{45}^{"} + A_{46}^{"} + A_{47}^{"}$$

$$= A_{48} - A_{49}^{sin}(\phi_{1c} + N_{31}^{}\phi_{T}) + A_{50}^{sin}(\gamma_{p}^{"} - \psi - \psi_{c})$$
(B176)

where B-5

$$A_{45} = I_3 A_{18} + I_{PR} A_{17} U - \frac{A_{15} A_{18}}{A_{44}} (I_2 N_{32} - \frac{A_{42}}{A_{31}} I_{1R} N_{31})$$
 (B177)

$$A_{46} = I_{PR}^{A}_{17}^{V} + A_{17}^{A}_{21}^{U^{2}} + \frac{A_{15}^{A}_{18}^{A}_{34}^{A}_{42}}{A_{31}^{A}_{44}} N_{31}^{2}$$
(B178)

$$A_{47} = 2 \frac{\omega^2}{|\omega|} A_{17} A_{20} U + \frac{A_{15} A_{18} A_{33} A_{42}}{A_{31} A_{44}} N_{31}$$
(B179)

 $^{^{}B-5}$ The coefficients ${\rm A}_{45}$ to ${\rm A}_{50}$ are only valid for entrance contact; i.e., for equation B100. If exit-coupled motion is in operation, equation B119 is applicable and ${\rm A}_{17}$ becomes ${\rm AA}_{17}$, and ${\rm A}_{18}$ becomes ${\rm AA}_{18}$.

$$A_{48} = \frac{-A_{15}A_{18}A_{32}A_{42}}{A_{31}A_{44}} - \frac{A_{15}A_{18}A_{43}}{A_{44}} T_2 - A_{16}A_{18}T_3 - \omega^2 A_{17}A_{19}$$
 (B180)

$$A_{49} = \frac{A_{15}^{A} 18^{A} 35^{A} 42}{A_{31}^{A} 44}$$
 (B181)

$$A_{50} = m_{\rm p} r_{\rm cp} {}^{\rm ft} 4^{\rm A}_{17} \omega^2$$
 (B182)

Contact Force Equations for Coupled Motion (applicable to both configurations)

The contact force F_{23} is given by equation B175:

$$F_{23} = \frac{1}{A_{44}} \left[\frac{-A_{32}A_{42}}{A_{31}} - T_{2}A_{43} - \frac{A_{33}A_{42}}{A_{31}} N_{31} - \frac{A_{42}A_{34}}{A_{31}} N_{31}^{2} \right]$$

$$- \frac{A_{35}A_{42}}{A_{31}} \sin(\phi_{1c} + N_{31}\phi_{T}) + (I_{2}N_{32} - \frac{A_{42}}{A_{31}} I_{1R}N_{31}) - \frac{A_{42}A_{31}}{A_{31}} \left[\frac{A_{42}A_{34}}{A_{31}} \right]$$
(B183)

The contact force F_{12} is found with the help of equation B171:

$$F_{12} = \frac{F_{23}^{A_{44}} + T_{2}^{A_{43}} - I_{2}^{N_{32}\phi}}{A_{42}}$$
 (B184)

The contact force P_n is given either by equation B96 or equation B117:

$$P_{n} = \frac{-I_{3}^{\phi} + F_{23}^{A}_{15} - T_{3}^{A}_{16}}{A_{17}}$$
 (B185)

Note that for exit-coupled motion; i.e., equation B117, the term A_{17} becomes AA_{17} . If this force is desired in terms of the pallet variable ψ , equation B76 or equation B107 must be used:

$$P_{n} = \frac{I_{PR} \ddot{\psi} + A_{21} \dot{\psi}^{2} + 2 \frac{\omega^{2}}{|\omega|} A_{20} \dot{\psi} + \omega^{2} A_{19} - m_{p} r_{cp} A_{2} \omega^{2} \sin(\gamma_{p}^{1} - \psi - \psi_{c})}{A_{18}}$$
(B186)

Again, note that for exit motion A_{18} becomes AA_{18} .

Differential Equations for Free Motion (applicable to both configurations)

Pallet Free Motion

By letting $P_n=0$ in equation Bl86, the single free motion differential equation for the pallet, which is now independent of exit or entrance contact, is obtained.

$$A_{51}\ddot{\psi} + A_{21}\ddot{\psi}^2 + A_{52}\dot{\psi} = -A_{53} + A_{54}\sin(\gamma_p^* - \psi - \psi_c)$$
 (B187)

where

$$A_{51} = I_{pR} \tag{B188}$$

$$A_{52} = 2 \frac{\omega^2}{|\omega|} A_{20}$$
 (B189)

$$A_{53} = \omega^2 A_{19}$$
 (B190)

$$A_{54} = \omega_{p}^2 r_{cp} R_4 \qquad (B191)$$

Escape Wheel - Gear Train - Rotor Free Motion

By setting $P_n=0$ in equation B96 or equation B117, the common differential equation of the system without the pallet is obtained (again independent of entrance or exit contact). This leads to:

$$^{"}_{3}^{\phi} = ^{A}_{15}^{F}_{23} - ^{T}_{3}^{A}_{16}$$
 (B192)

Equation B175 is then substituted into the above for F_{23} . The resulting differential equation is given by:

$$A_{55}^{\circ} + A_{56}^{\circ}^{\circ}^{2} + A_{57}^{\circ} = A_{58} - A_{59}^{\sin(\phi_{1c} + N_{31}^{\phi_{T}})}$$
 (B193)

where

$$A_{55} = I_3 - \frac{A_{15}}{A_{44}} N_{32}I_2 + \frac{A_{15}A_{42}}{A_{31}A_{44}} N_{31}I_{1R}$$
 (B194)

$$A_{56} = \frac{A_{15}A_{34}A_{42}}{A_{31}A_{44}} N_{31}^{2}$$
 (B195)

$$A_{57} = \frac{A_{15}A_{33}A_{42}}{A_{31}A_{44}} N_{31}$$
 (B196)

$$A_{58} = -A_{16}T_3 - \frac{A_{15}A_{43}}{A_{44}}T_2 - \frac{A_{15}A_{32}A_{42}}{A_{31}A_{44}}$$
 (B197)

$$A_{59} = \frac{A_{15}A_{35}A_{42}}{A_{31}A_{44}} \tag{B198}$$

Contact Force Equations for Free Motion (the subscript F stands for free motion) (applicable to both configurations)

Equation B192 may be solved for the free motion contact force F_{F23} , once ϕ is known:

$$F_{F23} = \frac{I_3^{\circ} + I_3^{\circ}_{16}}{A_{15}}$$
 (B199)

Equation B171, which was derived from the force and moment equations of gear and pinion set no. 2, may be modified to obtain the free motion contact force $\mathbf{F}_{\mathbf{F}12}$:

$$F_{F12} = \frac{F_{F23}A_{44} + T_{2}A_{43} - I_{2}N_{32}\phi}{A_{42}}$$
 (B200)

It must be understood that in both equations B199 and B200 the angular acceleration of free motion ϕ as obtained from the solution of equation B193 is used.

Impact

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While the following will show that the general form of the impact equation is the same for entrance and exit contact, it must be kept in mind that the expressions for the moment arms A_1^i , B_1^i , C_1^i and D_1^i depend on whether the angle α_{EX} or the angle α_{EX} is involved. (See equations D4, D5, D8, and D9 in appendix D.)

Entrance Impact

The general condition of entrance impact is illustrated in figure B-ll. As in reference B-2, it is assumed that the only acting impulse on the pallet and the escape wheel is represented by the mutual normal impulse P_n between these mechanism components. Experience has shown that the frictional impulse μP_n may be disregarded. Under these circumstances, the angular impulse on the pallet becomes:

$$\vec{J}_{p} = \vec{D}_{1} \vec{n}_{t} \times \vec{P}_{n} \vec{n}_{n} = \vec{P}_{n} \vec{D}_{1} \vec{k}$$
 (B201)

The angular impulse on the escape wheel is given by:

$$\vec{J}_3 = A_1^* \vec{n}_t \times (-P_n) \vec{n}_{ii} = -P_n A_1^* \vec{k}$$
 (B202)

If the above equations are made part of the usual angular momentum relationships, the components can be expressed as

$$I_{p}(\dot{\psi}_{f} - \dot{\psi}_{i}) = P_{p}D_{i}^{*}$$
 (B203)

and

$$I_3(\dot{\phi}_f - \dot{\phi}_1) = -P_nA_1^*$$
 (B204)

where I_p and I_3 represent the polar mass moments of inertia of the pallet and the escape wheel, respectively. The subscript i indicates the angular velocities before impact while f indicates the same quantities after impact. When P_n is simultaneously eliminated from equations B203 and B204, the following expression results:

$$I_{p}A_{l}^{\dagger}\dot{\psi}_{f} + I_{3}D_{l}^{\dagger}\dot{\phi}_{f} = I_{p}A_{l}^{\dagger}\dot{\psi}_{i} + I_{3}D_{l}^{\dagger}\dot{\phi}_{i}$$
 (B205)

Exit Impact

The conditions of exit impact are shown in figure B-12. With the same assumptions as in part a above, the expression for the angular impulse on the pallet becomes:

$$\vec{J}_{p} = D_{1}^{\prime} \vec{n}_{E} \times (-P_{n}) \vec{n}_{n} = -P_{n} D_{1}^{\prime} \vec{k}$$
 (B206)

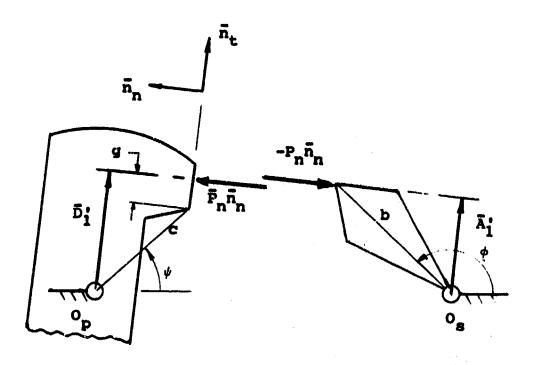


Figure B-11. Entrance impact

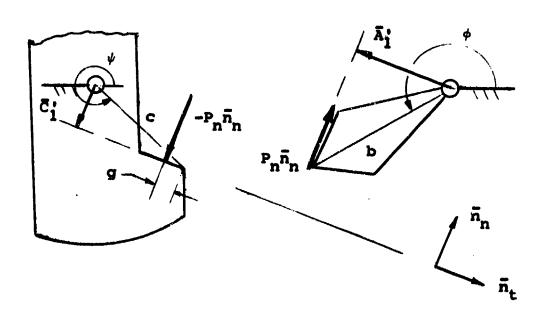


Figure B-12. Exit impact

The angular impulse on the escape wheel now becomes:

$$\vec{J}_3 = A_1^* \vec{n}_t \times (P_n) \vec{n}_n = P_n A_1^* \vec{k}$$
 (B207)

Again, these expressions are used in the angular momentum formulations of both components, as follows:

$$I_{p}(\mathring{\psi}_{f} - \mathring{\psi}_{i}) = -P_{n}D_{1}'$$
 (B208)

and

$$I_3(\dot{\phi}_f - \dot{\phi}_1) = P_n A_1^t$$
 (B209)

When P_n is now simultaneously eliminated from equations B208 and B209, equation B205 results again. Thus, identical formulations may be used for entrance and exit contact.

Impact Equations for Escapement

. In order to obtain expressions for the post impact angular velocities $\dot{\phi}_{\text{f}}$ and $\dot{\psi}_{\text{f}},$ the restitution equation

$$e_{r} = \frac{V_{TNf} - V_{SNf}}{V_{TN1} - V_{SN1}}$$
 (B210)

must be solved simultaneously with equation B205. V_{TNi} and V_{SNi} are defined by equations D13 and D15 in appendix D. The velocities V_{TNf} and V_{SNf} represent the normal contact point velocities after impact. After appropriate substitutions, the simultaneous solution of these expressions furnishes:

$$\dot{\phi}_{f} = \frac{\dot{\phi}_{1}(I_{3}D_{1}^{1^{2}} - e_{r}I_{p}A_{1}^{1^{2}}) + \dot{\psi}_{1}I_{p}A_{1}^{1}(1 + e_{r})}{I_{p}A_{1}^{1^{2}} + I_{3}D_{1}^{1^{2}}}$$
(B211)

and

$$\dot{\psi}_{f} = \frac{\dot{\phi}_{f} A_{1}' - e_{r} (\dot{\psi}_{1} D_{1}' - \dot{\phi}_{1} A_{1}')}{D_{1}'}$$
(B212)

Changes in Impact Equation due to Presence of the Rotor and Gear and Pinion No. 2

The presence of the rotor and gear and pinion no. 2 is accounted for by referring their moments of inertia to the escape wheel shaft. Thus, the total escape wheel moment of inertia is given by:

$$I_{STOT} = I_3 + I_2 N_{32}^2 + I_1 N_{31}^2$$
 (B213)

Equations B122 and B159 contain the above transmission ratios.

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REFERENCES

- B-1. C. G. Lowen and F. R. Tepper, "Computer Simulations of Artillery S&A Mechanisms (Involute Gear Train and Pin Pallet Runaway Escapement)," Technical Report ARLCD-TR-81039, ARRADCOM, Dover, NJ, July 1982.
- B-2. G. G. Lowen and F. R. Tepper, "Dynamics of the Pin Pallet Escapement," Technical Report ARLCD-TR-77062, ARRADCOM, Dover, NJ, June 1978.
- B-3. G. G. Lowen and F. R. Tepper, "Fuze Gear Train Analysis," Technical Report ARLCD-TR-79030, ARRADCOM, Dover, NJ, December 1979.

APPENDIX C KINEMATICS OF COUPLED MOTION

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Nomenclature

The following derivations for the kinematics of coupled motion are applicable to both entrance and exit motion. It must be remembered that for entrance motion, according to figure C-1:

$$\alpha = \alpha_{en}$$
 (C1)

and

while for exit motion:

$$\alpha = \alpha_{\alpha \nu}$$
 (C3)

and

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$$g = g_{ex}$$
 (C4)

Further, from figure C-1

- ϕ = Escape wheel angle, measured with respect to positive x' axis.
- ψ = Verge angle, measured with respect to positive x' axis. For entrance condition, defined by line 0 V. For exit condition, defined by line 0 W (measured ccw with respect to the positive x' axis).
- a = Distance between pivot points 0_p and 0_s
- b = Escape wheel radius
- c = For entrance condition, length O_pV . For exit condition, length O_pW (usually identical).

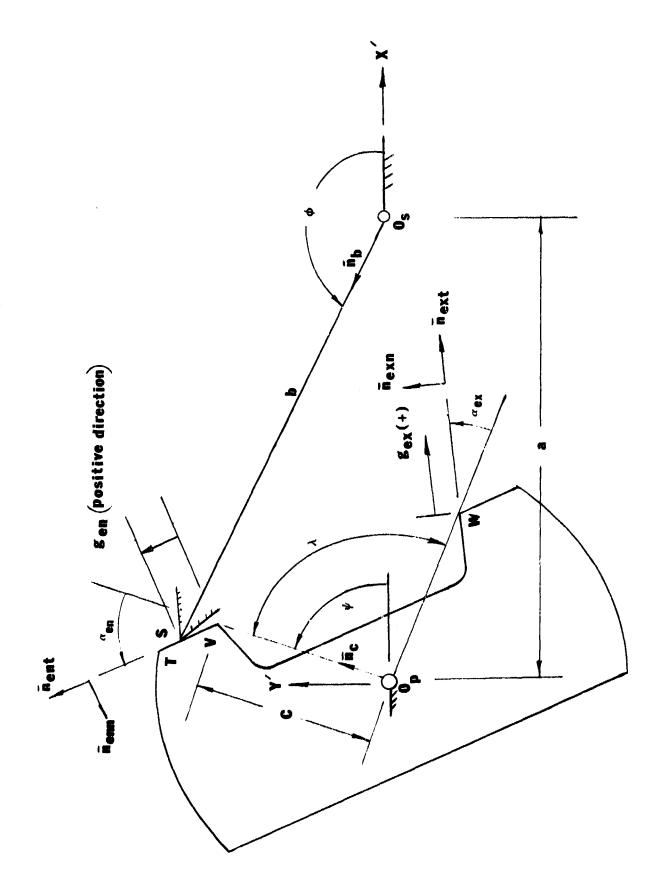
Unit Vectors

Tangential and normal unit vectors, which are associated both with the entrance and the exit working surfaces, are shown in figure C-1. When expressed without distinction between these surfaces, they become:

$$\vec{n}_{t} = \cos(\psi + \alpha)\vec{i}' + \sin(\psi + \alpha)\vec{j}'$$
 (C5)

and

$$\vec{n}_{n} = -\sin(\psi + \alpha)\vec{i}' + \cos(\psi + \alpha)\vec{j}'$$
 (C6)



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Figure C-1. Coupled motion nomenclature

$$\vec{n}_n = -\sin(\psi + \alpha)\vec{i}' + \cos(\psi + \alpha)\vec{j}'$$
 (C6)

Further, the unit vectors in directions $O_{\bf g}S$ and $O_{\bf p}V$ (or $O_{\bf p}W$), respectively, are given by

$$\vec{n}_b = \cos \phi \vec{i}^{\dagger} + \sin \phi \vec{j}^{\dagger} \qquad (C7)$$

and

$$\vec{n} = \cos \psi \, \vec{i}^{\dagger} + \sin \psi \, \vec{j}^{\dagger} \tag{C8}$$

Loop Equation and Input-Output Relationship

The loop equation for coupled motion is given below. Contact point S is at the tip of the escape wheel tooth. The coincident point on the pallet is point T (fig C-1).

$$\frac{\overrightarrow{O} \overrightarrow{V} + \overrightarrow{V} \overrightarrow{T} + \overrightarrow{SO}_{S} + \overrightarrow{O}_{S} \overrightarrow{O} = 0}{S p} = 0$$
 (C9)

Substitution of the appropriate dimensions and unit vectors into the above furnishes:

$$cn_c + gn_t - bn_b - ai' = 0$$
 (C10)

This becomes with the help of equations C5 to C8:

$$c(\cos\psi\vec{i}' + \sin\psi\vec{j}') + g(\cos(\psi + \alpha)\vec{i}' + \sin(\psi + \alpha)\vec{j}')$$

$$-b(\cos\phi\vec{i}' + \sin\phi\vec{j}') - a\vec{i}' = 0$$
(C11)

Appropriate ordering of equation C11 leads to component expressions which may be solved for g and ψ :

$$b\cos\phi + a - \cos\psi - g\cos(\psi + \alpha) = 0 \tag{C12}$$

and

$$b\sin\phi - c\sin\psi - g\sin(\psi + \alpha) = 0 \tag{C13}$$

Both of these component equations may be solved for g. Thus,

$$g = \frac{b\cos\phi + a - c\cos\psi}{\cos(\psi + \alpha)}$$
 (C14)

and

$$g = \frac{b\sin\phi - c\sin\psi}{\sin(\psi + \alpha)} \tag{C15}$$

g may be eliminated by setting equation C14 and C15 equal to each other. This leads to

$$(b\cos\phi + a - \cos\psi)\sin(\psi + \alpha) = (b\sin\phi - \cos(\psi + \alpha))\cos(\psi + \alpha)$$
 (C16)

After expansion and rearrangement, a four-bar linkage type expression is obtained which may be solved for the angle ψ according to the method shown by R. Hartenberg and J. Denavit.*

$$A\sin\psi + B\cos\psi = C \tag{C17}$$

where

$$A = a \cos \alpha + b \cos (\phi - \alpha)$$

$$B = a \sin \alpha - b \sin(\phi - \alpha)$$

 $C = c \sin \alpha$

The solution to equation Cl7 is of the form:

$$\psi = 2 \tan^{-1} \frac{A \pm \sqrt{A^2 + B^2 - C^2}}{B + C}$$
 (C18)

Once the correct ψ has been obtained for a given α or $\alpha_{\rm ex}$, the associated $g_{\rm en}$ or $g_{\rm ex}$ may be found with the help of equation C14 or equation C15. For the given directions of the associated unit vectors $g_{\rm en}$ will be a positive quantity while $g_{\rm ex}$ will be a negative one.

^{*} Kinematic Synthesis of Linkages, McGraw-Hill Publishing Co., New York, 1964.

Angular Velocity of Pallet

Implicit differentiation of equation C17 leads to the following expression for the angular velocity of the pallet when driven in coupled motion by the escape wheel:

$$\dot{\psi} = \dot{\phi} \frac{P}{O} \tag{C19}$$

where

 $\dot{\phi}$ = Angular velocity of the escape wheel

$$P = b\cos(\phi - \alpha - \varphi) \tag{C20}$$

$$Q = a\cos(\psi + \alpha) + b\cos(\phi - \alpha - \psi)$$
 (C21)

Relative Velocity of Contact Point S on the Escape Wheel with Respect to Contact Point T on the Pallet

With point S the tip of the escape wheel tooth and point T the coincident contact point of the pallet, the relative velocity $\overline{V}_{S/T}$ in the direction of the unit vector \overline{n}_t is found from (fig. C-1):

$$\vec{v}_{S/T} = [\vec{v}_{S/O_g} \cdot \vec{n}_t - \vec{v}_{T/O_p} \cdot \vec{n}_t] \vec{n}_t$$
 (C22)

The velocity $\overline{\mathbb{V}}_{S/O_{\boldsymbol{S}}}$ of point S with respect to the escape wheel pivot is obtained from

$$\overline{V}_{S/O_S} = \phi \overline{k} \times b \overline{n}_b$$
 (C23)

while $\overline{\mathtt{V}}_{\mathtt{T}/\mathtt{O}_p}$, the velocity of point T with respect to the pallet pivot, is given by:

$$\overline{V}_{T/O_{p}} = \sqrt[4]{k} \times (c\overline{n}_{c} + g\overline{n}_{t})$$
 (C24)

Substitution of equations C23 and C24, together with equations C5, C7, C8, into C22 leads, after the appropriate vector manipulations, to:

$$\vec{V}_{S/T} = -[\dot{\phi}b\sin(\phi - \psi - \alpha) + \dot{\psi}c\sin\alpha]\vec{n}_t$$
 (C25)

It can be shown that the above expression also represents the vector $\dot{g} = gn_{_{_{\rm I}}}$, as obtained from the rate of change of the quantity \bar{g} , which was defined by equations C14 and C15.

Angular Acceleration of the Pallet

Differentiation of equation C19 with respect to time results in the following expression for the pallet angular acceleration:

where, with equations C20 and C21

$$U = \frac{P}{Q} \tag{C27}$$

and

$$V = \frac{1}{Q^3} [aP^2 \sin(\psi + \alpha) - b(P - Q)^2 \sin(\phi - \alpha - \psi)]$$
 (C28)

APPENDIX D

MOMENT ARMS AND NORMAL VELOCITIES OF CONTACT POINTS

Moment Arms

The following derivations for the moment arm vectors, which find use in both coupled motion and impact calculations, are valid for both entrance and exit conditions (figs. D-1 and D-2) as long as the appropriate angle α or α is used. The variables ϕ and ψ must be available from the free motion computations.

The following loop for the determination of vectors \overline{A}_1^i and \overline{B}_1^i is shown by figure D-1:

$$A_1' \bar{n}_t + B_1' \bar{n}_n - b \bar{n}_b = 0$$
 (D1)

Substitution of the unit vectors given by equations C5 through C7 and subsequent separation of component expressions leads to the following two equations which may be solved for A_1^* and B_1^* :

$$A_{1}^{*}\cos(\psi + \alpha) - B_{1}^{*}\sin(\psi + \alpha) = b \cos \phi$$
 (D2)

and

$$A_1^* \sin(\psi + \alpha) + B_1^* \cos(\psi + \alpha) = b \sin \phi$$
 (D3)

Simultaneous solution of the above expressions for A_i^* and B_i^* leads to

$$\vec{A}_1^{\dagger} = b\cos(\phi - \psi - \alpha)\vec{n}_{t} \tag{D4}$$

and

$$\vec{B}_1' = bsin(\phi - \psi - \alpha)\vec{n}_n$$
 (D5)

Figure D-1 also serves for the determination of the vectors \bar{C}_1^* and \bar{D}_1^* . The loop equation

$$D_{1}^{*}\bar{n}_{t} + C_{1}^{*}\bar{n}_{n} - c\bar{n}_{c} - g\bar{n}_{t} = 0$$

furnishes, with the help of equations C5, C6, and C8 for the unit vectors, the following set of component expressions:

$$(D_1^* - g)\cos(\psi + \alpha) - C_1^*\sin(\psi + \alpha) = c \cos \psi$$
 (D6)

and

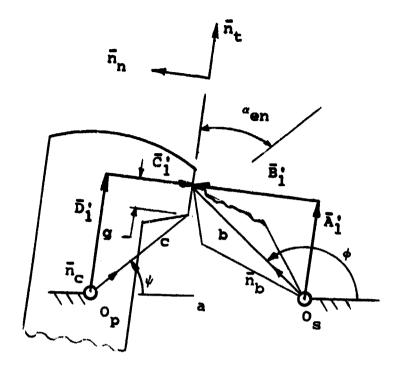


Figure D-1. Moment arms (entrance side)

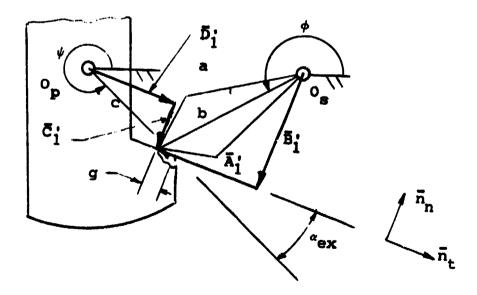


Figure D-2. Moment arms (exit side)

$$(D_i^t - g)\sin(\psi + \alpha) + C_i^t\cos(\psi + \alpha) = c \sin \psi$$
 (D7)

Simultaneous solution for C; and D; gives:

$$\overline{C}_{1}^{*} = -c\sin\alpha n \tag{D8}$$

and

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$$\bar{D}_{1}^{\prime} = (\cos\alpha + g)\bar{n}_{t} \tag{D9}$$

Again, it must be remembered to substitute the appropriate entrance or exit values for α and g.

Normal Velocities of Contact Points Before and After Impact

The contact point S on the escape wheel and the contact point T on the pallet are shown in figure D-3. At the instant before impact, the respective escape wheel and pallet angular velocities are given by:

$$\ddot{\phi}_{1} = \dot{\phi}_{1} \vec{k}' \tag{D10}$$

and

$$\dot{\psi}_{i} = \dot{\psi}_{i} \bar{k}' \tag{D11}$$

The associated velocity component $\overline{\textbf{V}}_{SN1}$ of the escape wheel contact point S is obtained from:

$$\tilde{V}_{SN1} = \tilde{\phi}_1 \times \tilde{A}_1' \tag{D12}$$

This velocity component is normal to the pallet face. This becomes, with the help of equations D10 and D4:

$$\vec{V}_{SN4} = b \dot{\phi}_4 \cos(\phi - \psi - \alpha) \vec{n}_{D}$$
 (D13)

In a similar manner the velocity component $\mathbf{v}_{\mathbf{TNi}}$ of the pallet contact point T becomes:

$$\vec{V}_{TN1} = \vec{\psi}_1 \times \vec{D}_1^* \tag{D14}$$

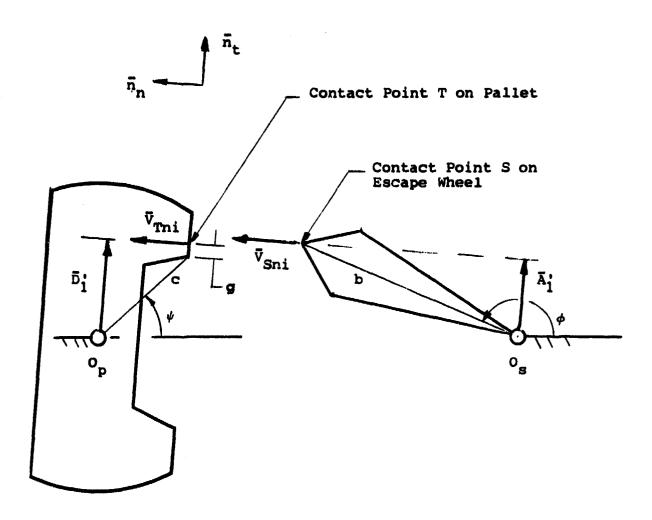


Figure D-3. Normal velocities of contact points S and T

This velocity component is also normal to the pallet face. Substitution of equations D9 and D11 into the above furnishes:

$$\bar{V}_{TNi} = \dot{\psi}_i (\cos\alpha + g)\bar{n}_n$$
 (D15)

When equations D13 and D15 are adapted to the post-impact normal velocities ∇_{SNf} and ∇_{TNf} , the respective angular velocities ϕ_f and ψ_f must be substituted (impact section of appendix B).

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APPENDIX E

KINEMATICS OF FREE MOTION: CONTACT SENSING EQUATIONS

The contact sensing quantities f and g' for entrance and exit conditions are defined by figures E-1 and E-2, respectively. It is assumed that the angles ϕ and ψ are known at every instant of free motion. To determine the entrance side sensing quantities, the following loop equation is written:

$$\frac{\overline{O_S} + \overline{ST} + \overline{TV} + \overline{VO_p} + \frac{\overline{O_O}}{\overline{P_S}} = 0$$
 (E1)

This becomes with the help of the appropriate unit vectors

$$b\vec{n}_b + f\vec{n}_n - g'\vec{n}_t - c\vec{n}_c + a\vec{i}' = 0$$
 (E2)

When a similar expression is written for the exit side (fig. E-2), the following relationship is obtained:

$$b\bar{n}_b - f\bar{n}_n + g'\bar{n}_t - c\bar{n}_c + a\bar{i}' = 0$$
 (E3)

The quantities f and g' have opposite signs in equations E2 and E3. If equation E2 is solved for f and g' with the help of the unit vectors of equations C5 to C8, then for the entrance side:

$$f = asin(\psi + \alpha) - bsin(\phi - \psi - \alpha) - csin\alpha$$
 (E4)

and

$$g' = a\cos(\psi + \alpha) + b\cos(\phi - \psi - \alpha) - \cos\alpha$$
 (E5)

The above expressions will yield positive numerical quantities. When the same expressions are used for the exit side (with the appropriate angle α), the resulting numerical quantities will be negative because of the aforementioned sign differences between equations E2 and E3.

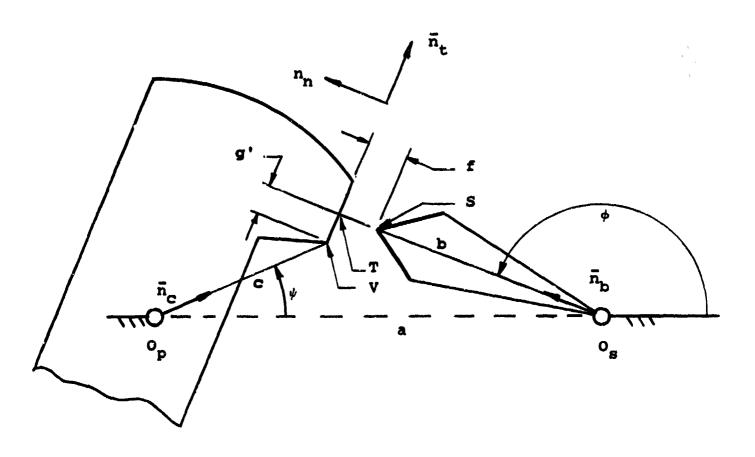


Figure E-1. Contact sensing on entrance side (center distance not to scale)

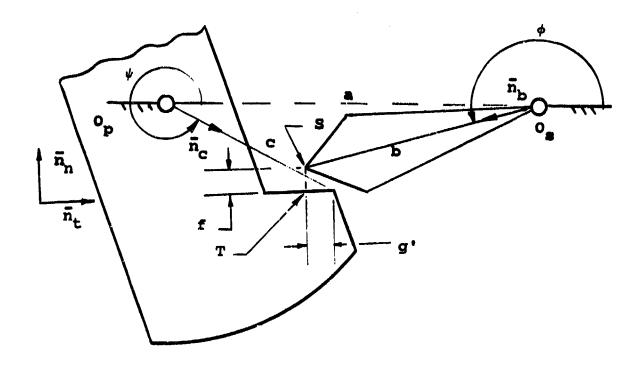


Figure E-2. Contact sensing on exit side (center distance not to scale)

APPENDIX F GEOMETRY OF STRAIGHT-SIDED VERGE

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Angles λ , α , and α ex

The nomenclature of the indicated straight sided verge is defined in figure F-1. The angle λ is obtained as follows:

$$\lambda = 2 \tan^{-1} \frac{h}{2L^{\dagger}} \tag{F1}$$

With the above, and angle η , the angles α and α become, respectively:

$$\alpha_{\rm en} = \pi - \frac{\lambda}{2}$$
 (F2)

$$\alpha_{\rm ex} = \frac{\lambda}{2} - \eta \tag{F3}$$

The angle η is a basic design dimension.

Lengths of Working Surfaces

Figure F-2 is used to find the distance $p_{\rm en}$; i.e., the length of the straight portion of the entrance working surface, as well as the corresponding length $p_{\rm ex}$ of the exit working surface of the verge.

The length pen = LV is determined from the difference in lengths

$$p_{en} = y(L') - \frac{h}{2}$$
 (F4)

where

y(L') = the y-coordinate of the point of intersection of the circle $x^2 + y^2 = r^2$, and the straight line x = L'

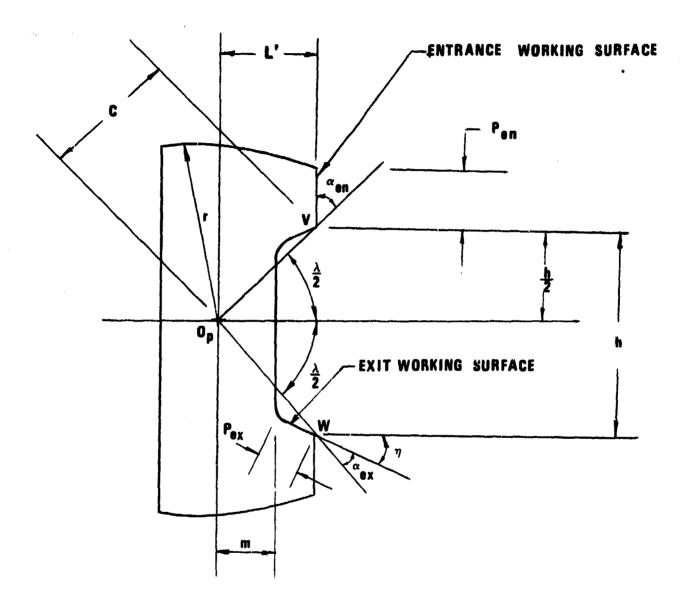
Then:

$$L^{+2} + y^2 = r^2$$

and

$$y = \sqrt{r^2 - L^2}$$

Equation F4 becomes:



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Figure F-1. Geometry of verge

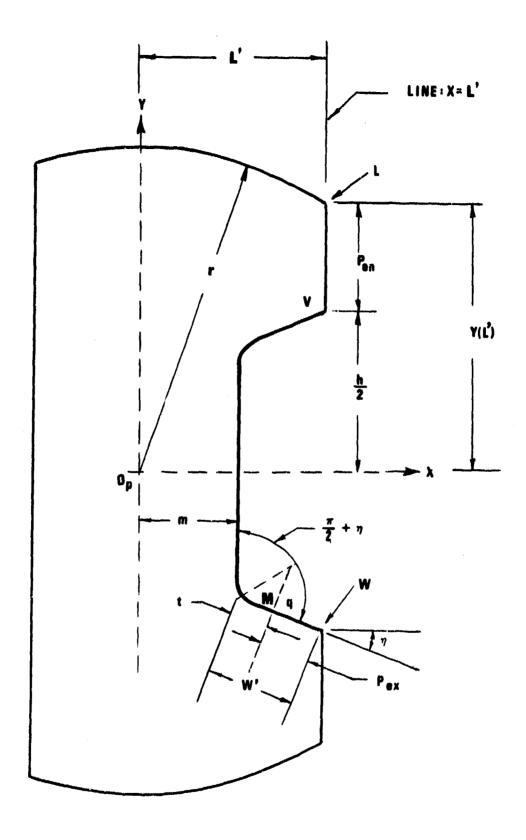


Figure F-2. Determination of p_{en} and p_{ex}

$$p_{en} = \sqrt{r^2 - L'^2 - \frac{h}{2}}$$
 (F5)

The length p_{ex} = MW is obtained with the help of

$$p_{ex} = W' - t (F6)$$

where point M represents the blend point between the fillet of radius q and the straight line of the exit working surface. The length W' is obtained from

$$W' = \frac{L' - m}{\cos \eta} \tag{F7}$$

while t is given by:

$$t = \frac{q}{\frac{\pi}{2} + \eta}$$

$$tan \left(\frac{\frac{\pi}{2} + \eta}{2} \right)$$
(F8)

where $\frac{\pi}{2} + \eta$ is the angle spanned by the fillet. With the above, equation F6 becomes

$$P_{\text{ex}} = \frac{L' - m}{\cos n} - \frac{q}{\frac{\pi}{2} + n}$$

$$\tan \left(\frac{\pi}{2}\right)$$
(F9)

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