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MEMORANDUM REPORT ARBRL-MR-03165

A COMPUTER CODE FOR THE SOLUTION OF THE
EQUATIONS GOVERNING A LAMINAR, PREMIXED,
ONE-DIMENSIONAL FLAME

Terence P. Coffee

April 1982



US ARMY ARMAMENT RESEARCH AND DEVELOPMENT COMMAND
BALLISTIC RESEARCH LABORATORY
ABERDEEN PROVING GROUND, MARYLAND

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REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER Memorandum Report ARBRL-MR-03165	2. GOVT ACCESSION NO. AD-A114041	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) A COMPUTER CODE FOR THE SOLUTION OF THE EQUATIONS GOVERNING A LAMINAR, PREMIXED, ONE-DIMENSIONAL FLAME	5. TYPE OF REPORT & PERIOD COVERED	
7. AUTHOR(s) Terence P. Coffee	6. PERFORMING ORG. REPORT NUMBER	
9. PERFORMING ORGANIZATION NAME AND ADDRESS US Army Ballistic Research Laboratory ATTN: DRDAR-BLI Aberdeen Proving Ground, MD 21005	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS 1L161102AH43	
11. CONTROLLING OFFICE NAME AND ADDRESS U.S. Army Armament Research & Development Command U.S. Army Ballistic Research Laboratory (DRDAR-BL) Aberdeen Proving Ground, MD 21005	12. REPORT DATE April 1982	
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)	13. NUMBER OF PAGES 141	
15. SECURITY CLASS. (of this report) Unclassified		
15a. DECLASSIFICATION/DOWNGRADING SCHEDULE		
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Laminar Flame Species Profiles One Dimensional Flame Temperature Profiles Premixed Flame Finite Elements Method Non-uniform Grid		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) (raj) A computer program for numerically solving the equations governing a laminar, premixed, one-dimensional flame is described. Both unbounded flames and burner stabilized flames are discussed. Some numerical considerations in successfully using the code are presented.		

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I. INTRODUCTION

We are interested in determining validated sets of elementary chemical reactions for use in predictive combustion models. For this purpose, a simulation of the laminar, premixed, one dimensional, steady state flame has been implemented. This has the advantage of being a relatively simple combustion problem that also yields predicted temperature and species profiles that can be compared with suitable burner experiments.

The basic idealism is of an infinite column of premixed gas. At some point the gas is ignited. A flame front forms and propagates along the column of gas. All effects are assumed to be one-dimensional. Eventually, the flame will reach steady state. A basic characteristic of the solution is the flame speed, the velocity with which the flame front moves with reference to the undisturbed gas mixture. The flame front is marked by steep temperature and chemical species profiles. The above formulation is referred to as an unbounded, or adiabatic flame.

In practice, a flame is usually stabilized on a burner. That is, the flame will propagate to the burner surface. Loss of heat to the burner will stabilize the flame. This creates different boundary conditions.

This report describes a method of numerically solving the equations governing a laminar, premixed, one-dimensional, steady state flame. Both unbounded flames and burner stabilized flames are discussed. To solve such systems, we have modified a standard package for integrating one-dimensional partial differential equations so that it efficiently handles flame equations. The basic procedure is to integrate the equations in time until the steady state solution is reached. Changes have been made in the basic structure of the code, as well as adding a number of subroutines. The program has been implemented for the ozone flame and the $H_2-O_2-N_2$ flame.

Section II describes the partial differential equations governing the flames. In Section III, the basic code used to solve the equations is introduced. Section IV describes the modifications made in the code, and Section V the additional subroutines added to the program. Section VI discusses the changes in the equations and the code necessary to model burner stabilized flames. Section VII reports some of the numerical considerations in successfully using the code. Finally, the Appendix gives a complete listing of the code, plus the output for a run modeling a typical $H_2-O_2-N_2$ flame.

II. THE UNBOUNDED FLAME EQUATIONS

The equations for a multicomponent reacting ideal gas mixture can be found in the literature.¹⁻⁴ We are interested in the equations that describe a one-dimensional, laminar, premixed flame that propagates in an unbounded medium. The effects of radiation, viscosity and body forces are ignored. The momentum equation

$$\hat{\rho} \frac{\partial \hat{v}}{\partial t} = - \hat{\rho} \hat{v} \frac{\partial \hat{v}}{\partial \hat{x}} - \frac{\partial \hat{p}}{\partial \hat{x}}$$

can be eliminated. Here \hat{v} is the fluid velocity, $\hat{\rho}$ is the density, and \hat{p} is the pressure. The independent variables are time t (sec) and distance \hat{x} (cm). For flames, the fluid velocity is much less than the speed of sound, that is, $v^2 \ll p/\rho$. For steady state flow, this implies that the pressure gradient is negligibly small. So the pressure is assumed to be constant.

The pertinent equations are then, overall continuity.

$$\frac{\partial \hat{\rho}}{\partial t} = - \frac{\partial (\hat{\rho} \hat{v})}{\partial \hat{x}} \quad . \quad (1)$$

That is, any change in density with respect to time is due to the overall convection of the mixture, which equals the gradient of the total mass flux $\hat{\rho} \hat{v}$.

Continuity of species:

$$\hat{\rho} \frac{\partial \hat{Y}_k}{\partial t} = - \hat{\rho} \hat{v} \frac{\partial \hat{Y}_k}{\partial \hat{x}} - \frac{\partial}{\partial \hat{x}} (\hat{\rho} \hat{Y}_k \hat{V}_k) + \hat{R}_k M_k, \quad k = 1, 2, \dots, N \quad . \quad (2)$$

¹F. A. Williams, Combustion Theory, Addison-Wesley, Reading, MA, Chapter 1, 1965.

²R. B. Bird, W. E. Stewart and E. N. Lightfoot, Transport Phenomena, John Wiley and Sons, NY, Chapter 18, 1960.

³R. M. Fristrom and A. A. Westenberg, Flame Structure, McGraw-Hill, NY, Chapter V-1, 1965

⁴J.O. Hirschfelder, C. F. Curtiss and R. B. Bird, Molecular Theory of Gases and Liquids, 2nd Printing corrected, with notes, John Wiley and Sons, NY, Chapter 11.1, 1964.

where \hat{Y}_k is the mass fraction of the k^{th} species, \hat{V}_k is its diffusion velocity, M_k is its molecular weight, and \hat{R}_k is the rate at which the species is produced or consumed by the chemical reactions. So changes in the concentration of the k^{th} species can be due to the convection of the species, the diffusion of the species in the mixture, or the production or consumption of the species by chemistry. The term $\rho \hat{Y}_k \hat{V}_k$ is the mass flux of the k^{th} species (relative to \hat{v}). Note that the total mass flux for the k^{th} species is $\hat{\rho}(\hat{v} + \hat{V}_k) \hat{Y}_k$.

Conservation of Energy.

$$\hat{\rho} \hat{c}_p \frac{\partial \hat{T}}{\partial t} = - \hat{\rho} \hat{v} \hat{c}_p \frac{\partial \hat{T}}{\partial x} + \frac{\partial}{\partial x} \left(\hat{\lambda} \frac{\partial \hat{T}}{\partial x} \right) - \sum_{k=1}^N \hat{R}_k M_k \hat{h}_k \\ - \hat{\rho} \sum_{k=1}^N \hat{c}_{pk} \hat{Y}_k \hat{V}_k \frac{\partial \hat{T}}{\partial x} \quad (3)$$

where \hat{T} is the temperature, \hat{c}_{pk} is the specific heat of the k^{th} species, \hat{c}_p is the specific heat of the mixture, $\hat{\lambda}$ is the thermal conductivity of the mixture, and \hat{h}_k is the specific enthalpy of the k^{th} species. So changes in the temperature can be due to the convection of heat, the conduction of heat, the production or consumption of energy by the chemical reactions, and a small amount due to the diffusion of species with different specific heats. We have not written a negligibly small term due to the species gradients (Dufour effect).

The thermal equation of state is given by the ideal gas law

$$\hat{p} = \hat{\rho} R \hat{T} \sum_{k=1}^N \hat{Y}_k / M_k, \quad (4)$$

where R is the gas constant. The caloric equation of state is

$$\hat{h}_k = \hat{h}_k^0 + \int_{\hat{T}_0}^{\hat{T}} \hat{c}_{pk} d \hat{T}, \quad (5)$$

where \hat{h}_k^0 is the specific enthalpy of the k^{th} species at some reference temperature \hat{T}_0 . The specific heat of the mixture is given by

$$\hat{c}_p = \sum_{k=1}^N \hat{c}_{pk} \hat{Y}_k. \quad (6)$$

From conservation of mass we have the relation

$$\sum_{k=1}^N \hat{Y}_k = 1 \quad (7)$$

and

$$\sum_{k=1}^N \hat{i}_k \hat{V}_k = 0. \quad (8)$$

The boundary conditions are the following. For $x = -\infty$.

$$\hat{T} = \hat{T}_U \text{ and } \hat{Y}_k = \hat{Y}_{kU}, \quad (k=1,2,\dots,N), \quad (9)$$

where \hat{T}_U is the temperature and the \hat{Y}_{kU} are the mass fractions of the original, undisturbed mixture. For $x = \infty$

$$\hat{T} = \hat{T}_B \text{ and } \hat{Y}_k = \hat{Y}_{kB}, \quad k=1,2,\dots,N. \quad (10)$$

where \hat{T}_B is the temperature and \hat{Y}_{kB} are the mass fractions of the burned mixture. \hat{T}_B is called the adiabatic temperature. Since we are assuming no heat loss to the surroundings, T_B and Y_{kB} depend only on the chemistry, and can be calculated in advance.

In practice, the integration will be over a finite interval (\hat{x}_L, \hat{x}_R) , where the flame front will be located roughly in the center of the interval. The boundary conditions (9) will be applied to \hat{x}_L . However, the conditions (10) are not convenient. After the flame front, there is a long recombination period before the mixture reaches adiabatic conditions. Choosing the interval of integration long enough to cover the entire recombination zone would be computationally expensive. So normally the flame will be cut off before it reaches adiabatic conditions,

and the weaker boundary conditions

$$\frac{\partial \hat{T}}{\partial \hat{x}} = \frac{\partial Y_k}{\partial \hat{x}} = 0, \quad k=1,2,\dots,N, \quad (11)$$

will be applied at \hat{x}_R .

At this stage the partial differential Eq. (1) for the density ρ can be eliminated by introducing a new coordinate ψ such that

$$\hat{\psi}(\hat{x}, \hat{t}) = \int_{\hat{x}_L}^{\hat{x}} \hat{\rho}(\hat{x}', \hat{t}) d\hat{x}' . \quad (12)$$

Then $\frac{\partial \hat{\psi}}{\partial \hat{x}} = \hat{\rho}$ and $\frac{\partial \hat{\psi}}{\partial \hat{t}} = -\hat{\rho}\hat{v} + \hat{m}_0(\hat{t})$, where $\hat{m}_0(\hat{t}) = \hat{\rho}\hat{v}|_{\hat{x}=0}$. With this

notation Eqs. (2) and (3) become

$$\tilde{\frac{\partial Y_k}{\partial t}} = -\tilde{m}_0 \frac{\partial \tilde{Y}_k}{\partial \tilde{\psi}} - \frac{\partial}{\partial \tilde{\psi}} (\tilde{P} \tilde{Y}_k \tilde{V}_k) + \tilde{R}_k \tilde{M}_k / \tilde{\rho} \quad (13)$$

and

$$\begin{aligned} \tilde{\frac{\partial T}{\partial t}} &= -\tilde{m}_0 \frac{\partial \tilde{T}}{\partial \tilde{\psi}} + \frac{1}{\tilde{c}_p} \left\{ \frac{\partial}{\partial \tilde{\psi}} \left(\tilde{\rho} \tilde{\lambda} \frac{\partial \tilde{T}}{\partial \tilde{\psi}} \right) \right. \\ &\quad \left. - \sum_{k=1}^N \tilde{R}_k \tilde{M}_k \tilde{h}_k / \tilde{\rho} - \sum_{k=1}^N \tilde{c}_{pk} \tilde{Y}_k \tilde{V}_k \frac{\partial \tilde{T}}{\partial \tilde{\psi}} \right\}, \end{aligned} \quad (14)$$

where the tilde variables are functions of $\tilde{t} = \hat{t}$ (sec) and $\tilde{\psi}$ (gm/cm^2).

For numerical convenience, dimensionless forms of Eqs. (13) and (14) are integrated. That is, we will define $T = \hat{T}/T_\infty$, $t = \hat{t}/t_\infty$ and $\psi = \hat{\psi}/\psi_\infty$, where t_∞ and ψ_∞ are chosen so as to obtain reasonable time and space scales for a given flame, and T_∞ is chosen so that T is the same order of magnitude as the larger Y_k 's. Then the equations are

$$\frac{\partial Y_k}{\partial t} = - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial Y_k}{\partial \psi} - \frac{t_\infty}{\psi_\infty} \frac{\partial}{\partial \psi} (\rho Y_k V_k) + t_\infty R_k M_k / \rho , \quad (15)$$

and

$$\begin{aligned} \frac{\partial T}{\partial t} = & - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial T}{\partial \psi} + \frac{t_\infty}{c_p} \left(\frac{1}{\psi_\infty^2} \frac{\partial}{\partial \psi} (\rho \lambda \frac{\partial T}{\partial \psi}) \right) \\ & - \sum_{k=1}^N R_k M_k h_k / (\rho T_\infty) - \frac{1}{\psi_\infty} \sum_{k=1}^N c_{pk} Y_k V_k \frac{\partial T}{\partial \psi} , \end{aligned} \quad (16)$$

where the variables are functions of t and ψ . The boundary conditions are

$$T = T_U , \quad Y_k = Y_{kU} , \quad k = 1, 2, \dots, N \quad (17)$$

at $\psi = \psi_L$ and

$$\frac{\partial T}{\partial \psi} = \frac{\partial Y_k}{\partial \psi} = 0 , \quad k = 1, 2, \dots, N \quad (18)$$

at $\psi = \psi_R$.

The integration can start from any initial profiles, as long as there is enough energy available to begin the combustion. As shown in Section V, the mass flux m_0 through the origin is modified as the integration proceeds so as to keep the flame front in the center of the interval of integration (ψ_L, ψ_R). The integration proceeds until m_0 approaches a constant, and the terms $\frac{\partial Y_k}{\partial t}$ and $\frac{\partial T}{\partial t}$ approach zero. The equation for species N is not integrated; Y_N is found from the relation (7).

The burning velocity, S , can be calculated at steady state. From the continuity equation $\partial(\rho v)/\partial \psi = 0$, or ρv is constant with respect to ψ . Then we can take any of the equations (15) and integrate over any interval (a, b) to obtain

$$\rho v_k [Y_k(b) - Y_k(a)] = \psi_\infty \int_a^b \rho^{-1} R_k M_k d\psi - \rho Y_k v_k |_{a^b}. \quad (19)$$

Then

$$v_k(-\infty) = \frac{\psi_\infty \int_a^b \rho^{-1} R_k M_k d\psi - \rho Y_k v_k |_{a^b}}{\rho(-\infty) [Y_k(b) - Y_k(a)]} \quad (20)$$

At steady state, all of the v_k are equal. Since in the present coordinate system the flame does not move, $v_k(-\infty)$ is the speed at which the unburned gas is approaching the flame. Conversely, $S_v = v_k(-\infty)$ is the speed at which the flame is propagating into the unburned mixture.

The parameters of the equations (15) and (16) are the specific heats c_{pk} , the specific enthalpies h_k , the chemistry production terms R_k , and the diffusion velocities v_k for each species, plus the thermal conductivity λ of the mixture. The c_{pk} and h_k are functions only of the temperature, and can be evaluated very accurately using sixth degree polynomial fits⁵. The chemistry is generally the least well known of the input data. We need to know which species are involved, which reactions can occur, and the rate constant for each reaction. The rate constants k_i will be of the form $a T^b \exp(c/T)$, where either b or c can be zero. We can then find the rate r_i for each reaction by multiplying the rate constant times the concentrations of the reactants (concentrations of the k^{th} species = $\rho Y_k / M_k$). Each R_k is then found by adding the rates of the reactions in which the k^{th} species is a reactant. The transport parameters λ and v_k are in general very complicated functions of temperature and species concentrations. Because of the numerical complexity, various approximations to these quantities have been used. We have the capability to run the code with a number of different levels of approximation. This is discussed in detail in another paper⁶.

⁵S. Gordon and B. J. McBride, "Computer Program for Calculation of Complex Chemical Equilibrium Compositions, Rocket Performance, Incident and Reflected Shocks, and Chapman-Jouguet Detonations", NASA-SP-273, 1971, (1976 program version).

⁶T. P. Coffee and J. M. Heimerl, "Transport Algorithms for Premixed, Laminar, Steady-State Flames", to be published in Combustion and Flame.

For future reference, we will discuss here the simplest transport algorithm used in the code. First assume that Ficks law,

$$\hat{Y}_k \hat{V}_k = - \hat{D}_{km} \frac{\partial \hat{Y}_k}{\partial \hat{x}} \quad (21)$$

is valid. This is only strictly true for a binary mixture where thermal diffusion is negligible. \hat{D}_{km} is a multicomponent diffusion coefficient. Equation (21) can be written as

$$\rho Y_k V_k = - \frac{\rho^2 D_{km}}{\psi_\infty} \frac{\partial Y_k}{\partial \psi}. \quad (22)$$

As a first approximation, we can take $\rho^2 D_{km}$ and $\rho \lambda$ to be constant. Also, we can assume that $c_p = c_{pk}$, $k=1,2,\dots,N$. Here c_p must be chosen so as to obtain the proper adiabatic temperature T_B . The procedure for choosing these constants and the justification of this approximation is given in reference 6. This level of approximation is reasonably accurate, if the constants are properly chosen.

Equations (15) and (16) now become

$$\frac{\partial Y_k}{\partial t} = - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial Y_k}{\partial \psi} + \frac{t_\infty}{\psi_\infty} \rho^2 D_{km} \frac{\partial^2 Y_k}{\partial \psi^2} + t_\infty R_k M_k / \rho \quad (23)$$

and

$$\frac{\partial T}{\partial t} = - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial T}{\partial \psi} + \frac{t_\infty}{c_p} \left\{ \frac{\rho \lambda}{\psi_\infty^2} \frac{\partial^2 T}{\partial \psi^2} - \sum_{k=1}^N \frac{R_k M_k h_k}{\rho T_\infty} \right\}. \quad (24)$$

Equation (5) now simplifies to

$$h_k = h_k^0 + c_p (T - T_0), \quad (25)$$

so the required input parameters for this level of approximation are the constants $\rho^2 D_{km}$, $\rho \lambda$, c_p , h_k^0 and the functions $R_k(T, Y_1, \dots, Y_N)$.

III. THE NUMERICAL METHOD - PDECOL

The package PDECOL, developed by Madsen and Sincovec⁷, was used to solve the equations. This package is designed to solve a general system of N nonlinear partial differential equations of at most second order on a finite interval. In our coordinate system the appropriate form is

$$\frac{\partial \vec{u}}{\partial t} = \vec{f}(t, \psi, \vec{u}, \vec{u}_\psi, \vec{u}_{\psi\psi}), \quad (26)$$

where

$$\begin{aligned}\vec{u} &= (Y_1, \dots, Y_{N-1}, T) \\ \vec{u}_\psi &= \left(\frac{\partial Y_1}{\partial \psi}, \dots, \frac{\partial Y_{N-1}}{\partial \psi}, \frac{\partial T}{\partial \psi} \right), \\ \vec{u}_{\psi\psi} &= \left(\frac{\partial^2 Y_1}{\partial \psi^2}, \dots, \frac{\partial^2 Y_{N-1}}{\partial \psi^2}, \frac{\partial^2 T}{\partial \psi^2} \right).\end{aligned}\quad (27)$$

Fairly general boundary conditions of the form

$$\vec{b}(\vec{u}, \vec{u}_\psi) = \vec{z}(t) \quad (28)$$

are allowed, where \vec{b} and \vec{z} are arbitrary vector valued functions with N components. Each solution component is assumed to be a known function of ψ at the initial time $t = t_0$. That is, $Y_k(t_0, \psi)$, $k=1,2,\dots,N-1$, and $T(t_0, \psi)$ are known functions. The initial conditions must be consistent with the boundary conditions.

The spatial discretization is accomplished by finite element collocation methods based on B-splines⁸. The user must supply a set of

⁷N. K. Madsen and R. F. Sincovec, "PDECOL: General Collocation Software for Partial Differential Equations", Preprint UCRL-78263 (Rev 1), Lawrence Livermore Laboratory, (1977).

⁸C. de Boor, "Package for Calculating with B-Splines", Siam, J. Numer. Anal. 14, 441-472, (1977).

NB breakpoints, that is, a set of strictly increasing locations where the polynomials are joined. He must also supply the order KORD of the splines and the number of continuity conditions, NCC, to be applied at the breakpoints. That is, if NCC=1, the approximating function is continuous; if NCC=2, it is continuous and smooth; and so on. PDECOL then generates a set of NC = KORD(NB-1) - NCC(NB-2) basis functions and collocation points. The basis functions $B_i(\psi)$ are piecewise polynomials of order KORD - 1. The basic assumption is that the solution can be written in the form

$$u_k = \sum_{i=1}^N c_k^{(i)}(t) B_i(\psi), \quad k=1, \dots, N, \quad (29)$$

where the basis functions $B_i(\psi)$ span the solution space for any fixed t to within a small error tolerance. The time dependent coefficients $c_k^{(i)}$ are determined uniquely by requiring that the expansions above satisfy the given boundary conditions and that they satisfy the partial differential equations exactly at the (N-2) interior (collocation) points. Since by definition a B-spline is zero except over a small interval, at any collocation point no more than KORD of the B-splines are non-zero. So the system of ODE's for the coefficients $c_k^{(i)}$ will not be fully coupled.

The boundary conditions (28) must also be changed into ordinary differential equations. This can be done by taking the derivative with respect to time, which results in

$$\sum_{j=1}^N \left[\frac{\partial b_k}{\partial u_j} \frac{\partial u_j}{\partial t} + \frac{\partial b_k}{\partial u_{j\psi}} \frac{\partial u_{j\psi}}{\partial t} \right] = \frac{dz_k}{dt}, \quad k=1, 2, \dots, N. \quad (30)$$

When equation (29) is substituted into equation (30), a set of ODE's in the $c_k^{(i)}$ results.

Unlike a finite difference code, the program can be run with no boundary condition at either the left or the right boundary. The program simply collocates at the boundary, using the same procedure as for the interior points. It must be, of course, the user's responsibility to define a mathematically meaningful PDE problem.

This system of ODE's is integrated in time, using a variant of the Gear stiff integrator⁹. This is a fully implicit, predictor-corrector method. The required banded Jacobian is generated internally by the program. Once the integrator has reached a desired output time, the values of Y_k and T can be obtained for any ψ by substituting into the expansions (29). The accuracy of the time integration is determined by a user supplied error tolerance ϵ .

For the actual integration, the code requires the values of the basis functions and the first two space derivatives only at the collocation points. Since computing these values is complicated, this is done once at the beginning of the program and the values are saved.

In general, a user must supply a main program and three subroutines. The program MAIN sets the values of the required parameters, calls the integrator, and writes any desired output. The subroutine UINIT gives the initial conditions. Given a collocation point ψ , the subroutine must return the vector $\vec{u}(t_o, \psi)$. The subroutine F evaluates the function f given by Eq. (26). Given t , ψ , u , u_ψ , and $u_{\psi\psi}$, the routine must return the vector $\partial u / \partial t$. The program automatically converts this to the corresponding set of ODE's involving the $c_k^{(i)}$. The subroutine BNDRY specifies the boundary conditions. That is, given t , ψ , \vec{u} , and \vec{u}_ψ , the subroutine returns the quantities $\partial b_k / \partial u_j$, $\partial b_k / \partial u_j$, $\partial z_k / \partial t$, $j, k = 1, 2, \dots, N$, for both ψ_L and ψ_R .

IV. MODIFICATIONS OF PDECOL

The time integration is controlled by a user supplied error tolerance ϵ . Single step error estimates divided by $C_{MAX_k}^{(i)}$ will be kept less than ϵ in the root-mean-square norm. In PDECOL, $C_{MAX_k}^{(i)}$ is initially set to the maximum of $|c_k^{(i)}|$ and 1.0. Thereafter, $C_{MAX_k}^{(i)}$ is the largest value of $c_k^{(i)}$ seen so far, or the initial $C_{MAX_k}^{(i)}$ if that is larger. However, this error criterion does not produce the desired accuracy for flame simulations, because radical species with small concentrations will control the flame, and must be computed accurately. But mass fractions much smaller than 1.0 will not be computed accurately using the original PDECOL criterion. An alternate criterion is the purely relative error criterion, that is, $C_{MAX_k}^{(i)} = |c_k^{(i)}|$. This criterion is also unacceptable because some species at some locations will approach zero, and excess computation will result

⁹A. C. Hindmarsh, "Preliminary Documentation of GEARIB: Solution of Implicit Systems of Ordinary Differential Equations with Banded Jacobian", Rep. UCID-30130, Lawrence Livermore Laboratory, (1976).

in accurately computing these negligible concentrations. Consequently a semi-relative error control is used. $C_{MAX_k}^{(i)}$ is chosen as the maximum of $c_k^{(i)}$ and a user supplied parameter SREC. Thus, mass fractions less than SREC will be computed less accurately. We have normally used $SREC=10^{-6}$.

The original program PDECOL is fully implicit. That is, it generates a set of $NO = N \times NC$ ordinary differential equations of the form

$$A \frac{d\vec{c}}{dt} = \vec{g}(t, \vec{c}), \quad (31)$$

where $\vec{c} = (c_1^{(1)} \dots c_N^{(1)}, c_1^{(2)} \dots c_N^{(2)} \dots)$. The resulting Jacobian $\frac{\partial \vec{g}}{\partial \vec{c}}$ is a banded NO by NO matrix. The band width is $3 \times ML+1$, where $ML = N (KORD - 1) - 1$. To advance a time step, the program first computes an explicit predictor for the values of the $c_k^{(i)}$ at the next time step. Then it solves a set of linear algebraic equations involving the Jacobian to correct the values. For a stiff system, such as one involving chemistry, this allows time steps orders of magnitude larger than those of an explicit method⁹. The drawback is the storage and execution time required to work with the Jacobian.

For example, consider a system where $KORD = 4$, $N = 10$, and $NC = 20$. The bandwidth of the Jacobian is 88, and we have a system of 300 ODE's. We then have 26,400 possible non-zero elements in the banded matrix. The amount of storage required also increases rapidly. For instance, if $N = 20$ instead of 10 the bandwidth will be 178, and we will require 106,800 words to store the Jacobian. Since we want to be able to solve systems at least this large, the storage requirements become almost prohibitive. In addition, solving such large systems of linear algebraic equations is very time consuming.

To avoid this problem, we essentially uncouple the partial differential equations and solve them successively. The basic procedure is illustrated below. Suppose we are at time t_n and we want to advance a time step to t_{n+1} . We first integrate the equation for Y_1 , under the assumption that Y_2, \dots, Y_{N-1}, T are constant at their t_n values. This uncouples the first PDE from the system. Subsequently, we integrate the second PDE for Y_2 , using the new value of Y_1 at t_{n+1} , and the old values for the other variables. Continuing this process, we finally solve for $T(t_{n+1})$, using updated values for all the mass fractions.

In general, this method is restricted to smaller step sizes than a fully implicit method. However, as steady state is approached, all the variables approach constants with respect to time, and the temporal coupling vanishes.

Now consider how this assumption affects the associated system (31) of ODE's. In the Jacobian, we will have

$$\partial g_k^{(i)} / \partial c_\ell^{(j)} = 0 \text{ if } k \neq \ell.$$

That is, changes in the ℓ^{th} PDE will not affect the k^{th} PDE if $k \neq \ell$. So most of the terms in the Jacobian will become zero.

The remaining nonzero elements are not in banded form. However, we can accomplish this by rearranging the vector $\vec{c} = (c_1^{(1)}, c_1^{(2)}, \dots, c_1^{(\text{NC})}, c_2^{(1)}, \dots, c_2^{(\text{NC})}, \dots)$. That is, the coefficients for each PDE are grouped together instead of the coefficients for each collocation point.

The Jacobian matrix can now be decomposed into N smaller $\text{NC} \times \text{NC}$ matrices on the main diagonal. Moreover, the band width of these matrices is only 3XML+1 , where $\text{ML} = \text{KORD} - 2$. The Jacobian is essentially N separate Jacobians, each for one PDE with NC collocation points.

The savings in storage space is dramatic. For our previous example of $\text{KORD} = 4$, $N = 10$, and $\text{NC} = 30$, we have 2100 nonzero elements instead of 26,400. If $N = 20$, we have only 4200 nonzero elements instead of 106,800.

The following procedure is used to actually integrate a time step. The predictor of the predictor-corrector method is used to obtain first estimates for all the $c_k^{(1)}$. Then the corrected $c_1^{(1)}$ are computed, using the first small Jacobian and assuming that the other $c_k^{(i)}$ do not change. Then the corrected values for $c_1^{(i)}$ and the predicted values for $c_k^{(i)}$, $k > 2$, are used, and the corrected $c_2^{(i)}$ are computed. This process continues through the N PDE's. The estimated error is calculated. If necessary, the above procedure is iterated. It is more efficient to use the predicted values rather than the values at the previous time step. In using the procedure, it is more efficient to integrate the minor species first, since they change most rapidly, then the major species, and finally the temperature.

The above procedure is similar to one developed by Spalding and Stephenson for use in a finite difference code¹⁰.

¹⁰D.B. Spalding and P.L. Stephenson, "Laminar Flame Propagation in Hydrogen + Bromine Mixtures", Proc. R. Soc. Lond. A. 324, 315-337 (1971).

PDECOL was rewritten to use the successive calculation method. This also means that we must change the user supplied routines F and BNDRY.

Because of the rearrangement of the vector \vec{c} , the core integrator will call F first at each collocation point, asking for the values of $\partial Y_1 / \partial t$. It will then repeat for each PDE. The routine F is written so as to return only the desired time derivative.

Since the problem can no longer directly handle coupling terms, the boundary conditions (28) must be uncoupled, that is, the boundary conditions must be of the form

$$b_k(u_k, u_{k\psi}) = Z_k(t) . \quad (32)$$

The rewritten subroutine BNDRY evaluates the quantities $\partial b_k / \partial u_k$, $\partial b_k / \partial u_{k\psi}$, $\partial Z_k / \partial t$, $k = 1, 2, \dots, N$.

Comparisons of the execution time of the fully implicit method versus the successive calculation method have not been made. However, our main purpose was not to reduce the execution time but to reduce the storage requirements. This iterative procedure accomplishes this goal and simultaneously gives accurate results in reasonable run times.

V. ADDITIONS TO PDECOL

The computer program PDECOL requires a set of user supplied subroutines. We have written a set of subroutines that casts the equations into a computationally efficient form and which generates the required output. These routines are the first seven listed in the appendix; namely MAIN, F, UINIT, FLSP, BNDRY, BKPT, and RT. In addition, several auxiliary programs are mentioned, but without providing a listing. These routines generate sets of input data or actual subroutines that are used frequently, or analyze the output of the flame code.

The code can be run with several different options. We first describe the case of an unbounded flame ($NBURN = 0$) using the simplest constant transport algorithm ($NTRAN = 1$). There is no information available about the flame speed or the solution profiles ($NSTART = 1$). Later in this section we describe restarting the integration ($NSTART = 2$), and using a more complicated transport algorithm ($NTRAN = 2$). Burner stabilized flames ($NBURN = 1$) are discussed in the next section.

Initially, a chemistry scheme must be chosen. In the appendix an $H_2 - O_2 - N_2$ system is used, with nine chemical species ($H, OH, O, HO_2, H_2O_2, H_2, O_2, H_2O, N_2$) and a set of thirty reactions involving these species, each with a rate constant of the form $a T^b \exp(c/T)$ (either b or c can be zero). This information is used as input to an auxiliary

code. This code writes the subroutine RT that computes the chemistry terms $R_k M_k / \rho$. The procedure is analogous to that used to write the transport subroutines⁶. This subroutine is attached to the flame code in the job stream and can be used for any problem involving this set of kinetics.

The actual subroutine RT has three main parts. The rate constants are evaluated for the current temperature and stored in the vector RK. Since the code uses a successive calculation method, the subroutine will be called N times at each time step and each collocation point. The rate constants are only recomputed if the temperature has been changed. Otherwise this section is skipped. Each rate constant is multiplied by the concentrations of the appropriate reactants divided by ρ to obtain r_j / ρ . The terms $R_k M_k / \rho$ (stored in the vector R) are calculated by adding the rates for the reactions in which Y_k is a product, subtracting the rates for which Y_k is a reactant, and multiplying by M_k .

The choice of a transport algorithm determines the form of the subroutine F. The version given in the appendix is for constant transport. Common statements are used to make the appropriate constants available for all subroutines. The chemistry terms required are found by calling RT.

The initial temperature T_U and mass fractions Y_{kU} of the unburned gas are input quantities. Another auxiliary code determines the adiabatic temperatures T_B and mass fractions Y_{kB} , and the constants $\rho^2 D_{km}$, h_k^o , $\rho\lambda$, and c_p needed by the constant transport algorithm (see reference 6). This information is saved on a data file and attached to TAPE 11 when the code is run.

The remaining data necessary to run the program is read in on cards (TAPE 5). The pressure p and the normalizing constants t_∞ , ψ_∞ , and T_∞ are specified. The option parameters NSTART, NTRAN, and NBURN are chosen. The numerical parameters required by the code are specified; that is, ψ_L and ψ_R , the final integration time t_{FINAL} , the time integration error control parameters ϵ and SREC, and a set of breakpoints.

The major difficulty in efficiently solving the flame equations is choosing an appropriate set of breakpoints. These must be close enough that spatial errors are minimized and yet not so dense that one's computer resources are exceeded. The breakpoints should be densest in the flame front, where the gradients are very steep.

The technique in the code is to use a static mesh, with the breakpoints most closely spaced near the center of the interval of integration. The flame front is then forced to remain near the center of the interval. This is done by adjusting m_0 , the mass flux through the origin. At steady state, the mass flux through the flame is a constant. So m_0 is

iteratively modified to match the steady state mass flow through the flame. This leads to a coordinate system in which the flame front is at rest. The transient behavior can cause the flame front to drift away from the center. This can also be corrected by modifying m_o .

This method requires a procedure for tracking the flame front. To do this the position of a specific temperature T_{cn} is monitored. The code attempts to keep this temperature located at the center of the interval of integration. As a heuristic rule this temperature is defined by

$$T_{cn} = T_U + 0.4 (T_B - T_U) . \quad (33)$$

The average of T_U and T_B is not used. This is because the mixture will not normally reach the adiabatic temperature at the end of the flame front. Rather, there is a radical overshoot, and the recombination of these radicals will very slowly raise the temperature. T_{cn} as defined will usually be close to the center of the flame front. The details of this iterative procedure are discussed in reference 11.

So the breakpoints may be chosen to be densest in the center of the interval of integration. However, it is tedious to have to choose an entire breakpoint sequence for each problem. We have developed a procedure to generate an appropriate type of breakpoint sequence from a small number of parameters. By varying three parameters, a wide variety of breakpoint sequences can be generated.

The user must supply NINT, the number of intervals ($NB = NINT+1$), NCN, the number of intervals of equal length that will be at the center of the interval, and FC, the ratio between the longest intervals (on the boundaries) and the shortest intervals. Also let L be the total length of the interval of integration, that is, $L = \psi_R - \psi_L$. The program generates a set of intervals whose lengths increase by a constant factor α , where

$$\alpha = \log^{-1} [2(\log FC)/(NINT-NCN)] . \quad (34)$$

The common length LC of the NCN center shortest intervals is

$$LC = L/[NCN + 2 \alpha (\alpha^{(NINT-NCN)/2} - 1)/(\alpha - 1)]. \quad (35)$$

The procedure can best be seen by example. Suppose we have $\psi_L = 0$,

¹¹T.P. Coffee and J.M. Heimerl, "A Method for Computing the Flame Speed of a Laminar, Premixed, One Dimensional Flame", BRL Technical Report ARBRL-TR-02212, January 1980.

$\psi_R = 10$, $NINT = 12$, $NCN = 4$, and $FC = 6$. Then $L = 10$, $\alpha = 1.5651$, and $LC = .3155$. The resulting breakpoint sequence is given in Table 1. Note that the 4 center intervals are of the same length, the length of the intervals then increases by a factor of α , and the two intervals by the boundaries are 6 times as long as the central intervals. So the procedure automatically generates a set of breakpoints that are closest together near the center, with the spacing increasing smoothly toward the boundaries.

Some experimentation is necessary to choose the proper value of the above parameters. However, choosing $NINT = 12$, $NCN = 4$, and FC between 4 and 8 has worked in most of the cases we have tried. Normally we experiment using the constant transport algorithm, and then use a more realistic transport subroutine once we have a good breakpoint sequence.

TABLE 1. THE SET OF BREAKPOINTS GENERATED BY $L = 10$, $NINT = 12$, $NCN = 4$ and $FC = 6$.

<u>Breakpoints</u>	<u>Interval Lengths</u>
0.0	
1.8929	1.8929
3.1024	1.2094
3.8752	.7728
4.3689	.4937
4.6844	.3155
5.0000	.3155
5.3155	.3155
5.6310	.3155
6.1247	.4937
6.8975	.7728
8.1069	1.2094
10.0000	1.8929

The breakpoint sequence is written by the subroutine BKPT. It also generates a larger set of evaluation points by interpolating between the breakpoints. Using Eq. (29), the subroutine VALUES can evaluate the Y_k and T at this larger set of points. This information is useful in generating detailed output, such as graphs, or in performing numerical integrations (see below).

The subroutine UINIT writes the initial profiles, where

$$\begin{aligned}\psi_1 &= \psi_L + 0.24 (\psi_R - \psi_L) \\ \psi_2 &= \psi_L + 0.64 (\psi_R - \psi_L)\end{aligned}\quad (36)$$

and

$$Y_k(t_0, \psi) = \begin{cases} Y_{kU}, & \psi_L \leq \psi \leq \psi_1. \\ Y_{kU} + (Y_{kB} - Y_{kU}) \sin \left[\frac{\pi}{2} \left(\frac{\psi - \psi_1}{\psi_2 - \psi_1} \right)^2 \right], & \psi_1 \leq \psi \leq \psi_2 \\ Y_{kB}, & \psi_2 \leq \psi \leq \psi_R. \end{cases} \quad (37)$$

The temperature T is defined similarly.

This particular definition will give us $T = T_{cn}$ at $\psi_{cn} = 0.5(\psi_L + \psi_R)$. The choice of the particular function that defines the Y_k and T between ψ_1 and ψ_2 is not important. We have used a straight line with success, but defining a smooth function is slightly more efficient.

A requirement of PDECOL is that the initial profiles satisfy the boundary conditions, in our case given by Eqs. (17) and (18). The above initial profiles have the proper unburned values at ψ_L and are constant (space derivative zero) near ψ_R .

For this case the subroutine BNDRY has a simple form. At ψ_L , the subroutine returns the values $\partial b_k / \partial u_k = 1$, $\partial b_k / \partial u_{k\psi} = 0$, and $\partial Z_k / \partial t = 0$. At ψ_R , the conditions are $\partial b_k / \partial u_j = 0$, $\partial b_k / \partial u_{k\psi} = 1$, and $\partial Z_k / \partial t = 0$.

To begin the integration, a starting value for m_0 , the mass flux through the origin is required. To do this, the code ignores the time dependent terms, assumes that the mass flux ρv is constant, and uses Eq. (20) to obtain a value of ρv . This gives a reasonable starting value for m_0 .

The evaluation of Eq. (20) is carried out in the subroutine FLSP. The integral is approximated using the trapezoidal rule, where the integrand $R_k M_k / \rho$ is calculated at the evaluation points. The flame speed $v_k(-\infty)$ is evaluated for each species k and for several intervals (a, b) , where $a = \psi_L$. The initial value for m_0 is based on species N-1 integrated over the entire interval (ψ_L, ψ_R) .

The integration is performed over several intervals (a, b) because of the behavior of the minor species. These are normally close to zero at ψ_L , reach a peak in the flame front, and are close to zero again at ψ_R . For these species, integrating from the left boundary to the flame front is much more accurate.

The chemistry terms in Eq. (20) are found by calling RT. The diffusion terms for this case are found using Ficks law, Eq. (22). Since we are assuming that $\rho^2 D_{km}$ is constant, this value is stored in the vector R2D. The subroutine FLSP also computes the x values from the relation

$$x(t, \psi) = \psi_\infty \int_{\psi_L}^{\psi} \rho^{-1} d\psi, \quad (38)$$

using the trapezoidal rule, and computes an estimate of the flame thickness.

As the time integration proceeds, control returns to MAIN at a series of output times. The present spatial location of the temperature T_{cn} is found, and this is used to find the average value of the mass flux since the last output time. The function $m_0(t)$ is redefined at these times so as to keep the flame front in roughly the same position. The output times are chosen by the program so that the flame front will not drift too far between evaluations. Also FLSP is called so the user can see if the flame speeds computed from the different species profiles are approaching a common value.

At the final time t_{FINAL} , FLSP also writes an output file. It consists of all the evaluation points ψ_i , the corresponding x_i , the

values of Y_k and T , plus their first and second derivatives with respect to ψ . This file can be attached to an output routine. By also attaching F and RT , we can compute and print out any quantity in the steady state solution in which we are interested. This file can also be attached to a graphics routine.

Similarly, the program MAIN writes a restart file. This consists of the present location of T_{cn} , the present value of m_0 , and the values of Y_k and T at the collocation points. This file can be used to restart the time integration. It is attached to TAPE1, and the parameter NSTART is set equal to 2. The input parameters read in on cards can now be changed if desired. UINIT will translate these input values to center the flame front, and will use interpolation to find the appropriate values of the starting profiles at the new collocation points. The old value of m_0 is used to start the integration.

So far only the constant transport case ($NTRAN = 1$) has been considered. For more realistic algorithms F is written by an auxiliary code and attached to the flame code ($NTRAN = 2$). Several different levels of approximation can be used.⁶

In these more complicated algorithms, the diffusion velocities V_k are coupled, and they must be computed simultaneously. But since the code uses successive calculation, only the value of one V_k is required on each call to F . Because the computation is time-consuming, it is preferable not to compute the V_k N times at each time step for each collocation point. To economize computer time, all the thermodynamic and transport quantities required at all the collocation points for $k = 1$ are computed, and stored in vectors. For $k > 1$, we use the same values, even though some of the Y_k terms have changed slightly. Only the chemistry terms are reevaluated. Since the chemistry normally changes much more rapidly than the transport, this will still be a good approximation.

In generating an approximation to the Jacobian (using finite differences) it is necessary to recompute the transport each time F is called, since what is of interest is the effect of changes in Y_k and T on the time derivatives. Ignoring the changes in transport leads to an inaccurate Jacobian. However, the rate constants k_j and the thermodynamic quantities c_p and h_k need be recomputed only if T is changed, since they only depend on temperature.

As will be seen in the next section it will still be useful to formulate the mass flux in a Ficks law form Eq. (22). For the more complicated transport subroutines, we define

$$\rho^2 D_{km} = - \frac{\psi_\infty \rho Y_k V_k}{\partial Y_k / \partial \psi} . \quad (39)$$

This new quantity $\rho^2 D_{km}$ is no longer constant with respect to space or time.

VI. BURNER STABILIZED FLAMES

So far we have only discussed flames that propagate in an unbounded medium. In actual experiments, the flame will usually be stabilized by a burner. It is useful to be able to model this type of experiment so that we can compare experimental and calculated profiles.

Our basic idealization is of a cylindrical, porous plug burner. The premixed gas exits the plug with a constant velocity v , but the plug prevents back diffusion of the products into the burner. When the gas is ignited, a flame front develops and propagates toward the burner. The gas velocity v at the burner must be less than the flame velocity, or the flame will be blown off the burner. In the model this will look like an unbounded flame. In an experiment the flame will be extinguished by the surrounding atmosphere. As the flame approaches the burner, the burner surface acts as a heat sink for the flame. The loss of heat slows down the flame velocity, until the flame stabilizes near the surface of the burner at the gas velocity. As the gas velocity v is decreased, the flame loses more heat to the burner, and stabilizes closer to the burner surface. If v is made too small in an experiment, the flame can flash back into the burner.

For this kind of burner, air is entrained along the outside edges of the flame. However, the center of the flame will correspond closely to a premixed, laminar, one-dimensional flame.

The only change in the equations is in the left boundary condition at the surface of the burner. Because of back diffusion from the flame to the burner surface, the mass fractions of the unburned mixture are not conserved. However, due to conservation of mass, the mass flux fractions, defined as

$$\epsilon_k = \frac{\rho Y_k v + \rho Y_k V_k}{\rho v} = Y_k + \frac{\rho Y_k V_k}{\rho v} \quad (40)$$

are conserved. Within the burner, the diffusion velocities are zero, and $\epsilon_{kU} = Y_{kU}$. So the appropriate boundary conditions at the burner surface are

$$\epsilon_k = Y_{kU}. \quad (41)$$

The boundary conditions for the temperature equation depends on how heat

is extracted. We assume that the burner is maintained at a constant temperature. Then the boundary condition is the same as for an unbounded flame,

$$T = T_U. \quad (42)$$

This idealization is discussed by Hirshfelder, Curtiss and Bird¹².

To implement this in the flame code, the input parameter NBURN is set equal to 1. The fluid velocity at the burner must be specified. Then m^0 is a predetermined constant instead of an adjustable parameter. All the other input data remains the same. However, the code will handle this data differently.

The breakpoints are generated differently, since the flame front will be near the left boundary instead of in the center of the interval of integration. NCN is the number of intervals of equal length at the left boundary. FC is the ratio between the longest interval (at the right boundary) and the shortest interval (at the left boundary). The constant α is now defined by

$$\alpha = \log^{-1} [(\log FC)/(NINT-NCN)]. \quad (43)$$

The common length LC of the NCN shortest intervals is now

$$LC = L/[NCN + \alpha(\alpha^{(NINT-NCN)} - 1)/(\alpha - 1)]. \quad (44)$$

As an example, consider the same input parameters that we used for an unbounded flame; $\psi_L = 0$, $\psi_R = 10$, NINT = 12, NCN = 4, and FC = 6.

Then $L = 10$, $\alpha = 1.2510$ and $LC = .4181$. The resulting breakpoint sequence is given in Table 2. The shortest intervals are near the burner surface, where fairly steep gradients are expected.

In general, it is possible to choose a shorter interval of integration L for a burner stabilized flame. There is no longer a need for a relatively long interval to the left of the flame front in order to approach the unburned conditions. The number of breakpoints can then also be reduced.

The initial profiles are also chosen differently. The code sets

¹²J.O. Hirshfelder, C.F. Curtiss, and R.B. Bird, op. cit., pp. 761-763.

TABLE 2. THE SET OF BREAKPOINTS GENERATED BY L=10, NINT = 12,
NCN = 4 and FC = 6 (NBURN = 1)

<u>Breakpoints</u>	<u>Interval Lengths</u>
0	.3458
.3458	.3458
.6916	.3458
1.0374	.3458
1.3832	.4326
1.8159	.5412
2.3571	.6771
3.0342	.8471
3.8812	1.0597
4.9409	1.3257
6.2666	1.6585
7.9251	2.0749
10.0000	

$$\psi_1 = \psi_L + 0.1 (\psi_R - \psi_L) \quad (45)$$

$$\psi_2 = \psi_L + 0.3 (\psi_R - \psi_L)$$

and then uses Eq. (37) as before. This flame front is more likely to be close to the position of the final steady state stabilized flame. Also, the flame front is chosen to be narrower. If Eq. (36) is used, the initial transient flame velocity may be less than the velocity of the fluid. The flame then drifts toward the right, and can be carried completely outside the interval of integration before it stabilizes. Choosing a steeper flame front leads to a larger initial flame velocity.

The boundary condition, Eq. (41), can cause some difficulty. It is fairly straightforward for the constant transport case. Using Fick's law, Eq. (22), it can be rewritten as

$$Y_k - \left(\frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial Y_k}{\partial \psi} \right) \Big|_{\psi_L} = Y_U . \quad (46)$$

Recall that PDECOL converts this to a time dependent equation by taking the time derivative. In the form used by PDECOL, Eq. (46) becomes

$$\left[\frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial}{\partial t} \left(\frac{\partial Y_k}{\partial \psi} \right) \right] \Big|_{\psi_L} = 0 . \quad (47)$$

So the subroutine BNDRY must now return the values $\partial b_k / \partial u_k = 1$, $\partial b_k / \partial u_k = -\rho^2 D_{km} / \psi_\infty m_0$, and $\partial Z_k / \partial t = 0$ at ψ_L for $k = 1, 2, \dots, N-1$. The mass flux fractions will then have the proper values as the integration proceeds.

For more realistic transport algorithms the mass flux $\rho Y_k V_k$ is a complicated function of all the mass fractions. But by using Eq. (39) to define $\rho^2 D_{km}$, the boundary condition can be put in the same form as Eq. (46), except that $\rho^2 D_{km}$ is not a constant. Taking the time derivative we obtain

$$\left\{ \frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial}{\partial t} \left(\frac{\partial Y_k}{\partial \psi} \right) - \frac{1}{\psi_\infty m_0} \frac{\partial}{\partial t} (\rho^2 D_{km}) \frac{\partial Y_k}{\partial \psi} \right\} \Big|_{\psi_L} = 0 . \quad (48)$$

The time derivative of $\rho^2 D_{km}$ can not be evaluated. It is not possible to write it as an explicit function of the Y_k and $\partial Y_k / \partial \psi$. Moreover, since the code uses successive calculation, cross coupling terms between the different species are not allowed. By necessity, the last term in Eq. (48) must be ignored.

It is possible to just use the expression

$$\left\{ \frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial}{\partial t} \left(\frac{\partial Y_k}{\partial \psi} \right) \right\} \Big|_{\psi_L} = 0 \quad (49)$$

as the boundary condition. At steady state, the mass flux fractions will approach constants. But because of the transient behavior of $\rho^2 D_{km}$,

the mass flux fractions will converge to incorrect values.

What is required is a correction term that will cause the code to converge to the proper steady state limit. Moreover, it must be of a form that can be used in the code. This can be done by choosing the boundary condition

$$\left\{ \frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial}{\partial t} (\frac{\partial v_k}{\partial \psi}) \right\} |_{\psi_L} = \frac{\partial Z}{\partial t}, \quad (50)$$

where

$$\frac{\partial Z}{\partial t} = \frac{Y_{kU} - \epsilon_k|_{\psi_L}}{0.1 t_{FINAL}}. \quad (51)$$

The rationale is that $\rho^2 D_{km}$ changes rather slowly. As it changes the mass flux fraction at ψ_L will change slowly from the proper value. We use a heuristically defined function $\partial Z / \partial t$ to modify the value of ϵ_k until it again equals Y_k . So as the time integration proceeds, the value of ϵ_k at ψ_L will vary slightly, but will approach the proper steady state limit.

It is necessary to check that the boundary conditions are still consistent with the initial conditions. For the initial profiles defined by Eq. (37) this will be the case. The mass fractions Y_k are constant near ψ_L , so $\partial Y_k / \partial \psi$ at ψ_L will be zero. Also $\partial T / \partial \psi$ at ψ_L is zero. For any transport algorithm all the V_k will be zero at ψ_L , and $\epsilon_k = Y_k$. So the initial profile is consistent with the boundary conditions.

When the integration is restarted (NSTART = 2), the space derivatives at ψ_L are normally non-zero. It is necessary to modify the way PDECOL determines the initial values of $c_k^{(i)}$. Normally this is done by the subroutine INITAL. It calls the user supplied subroutine UINIT to obtain the values $u_k(t_0, \psi_j)$, where the ψ_j , $j = 1, 2, \dots, NC$ are the collocation points. Then by substituting into the expansion (29), it obtains a set of linear algebraic equations of the form

$$u_k(t_0, \psi_j) = \sum_{i=1}^{NC} c_k^{(i)}(t_0) B_i(\psi_j) \quad k = 1, 2, \dots, N, \quad j = 1, 2, \dots, NC. \quad (52)$$

These systems of equations are solved to obtain the initial $c_k^{(i)}(t_0)$.

For a burner stabilized flame, the appropriate values of the mass fractions at $\psi_1 = \psi_L$ are not known but the values of the mass flux fractions ϵ_k are. Thus, we use the boundary condition (46), substitute the expansion (29), and obtain

$$Y_{kU} = \sum_{i=1}^{NC} c_k^{(i)}(t_0) \left[B_i(\psi_1) - \frac{\rho^2 D_{km}}{\psi_\infty} - \frac{\partial B_i(\psi_1)}{\partial \psi} \right], \quad k=1,2\dots N-1. \quad (53)$$

The values of $\rho^2 D_{km}$ from the last run of the code are used. These values are also saved in the restart file. So for a burner stabilized flame, the subroutine INITIAL has been changed so that it generates this new set of equations for $j = 1, k = 1,2\dots N-1$.

Note that the above procedure is not strictly necessary. Since we have a correction term $\partial Z / \partial t$, the boundary conditions will approach the proper values, even if they are incorrect at the start of the integration. But it is more efficient to begin with the boundary conditions as accurate as possible.

To actually make comparisons with experiments, some of the boundary conditions may have to be further modified. For instance, evidence exists that hydrogen atoms H combine very rapidly on the burner surface to form H_2 .¹³ For practical purposes, we consider this to be instantaneous. Then if our first species is H and our second species is H_2 , their boundary conditions are

$$Y_1 = 0 \quad \text{at } \psi_L \quad (54)$$

and

$$\epsilon_1 + \epsilon_2 = Y_{2u} \quad \text{at } \psi_L. \quad (55)$$

These changes can easily be made in UINIT and INITIAL.

As the time integration proceeds, control returns to the routine MAIN at five equally spaced output times. The code calls subroutine FLSP so the integration process can be monitored. There is no need to adjust m_0 , since this is now a predetermined constant.

¹³J. Warnatz, "Calculation of the Structure of Laminar Flat Flames III: Structure of Burner-Stabilized Hydrogen-Oxygen and Hydrogen-Fluorine Flames", Verlog Chemie GmbH, BdB 8/78 E 4018.

VII. NUMERICAL CONSIDERATIONS

This code has been applied to the H₂-O₂-N₂ system, with nine chemical species. An earlier version of the code has been applied to the ozone system, with three species¹⁴. Several calculations have been performed for methane-air flames. In each case the code integrates in time until the steady state solution has been reached. What is desired is accurate values for the flame speed and accurate temperature and species profiles. Results are not discussed here, since they are given in the papers referenced above. Instead, we will discuss some of the considerations necessary to obtain accurate results, based on our experiences with the above systems.

The flame speed can be calculated from any of the species profiles using Eq. (20). Normally the time integration proceeds until the flame speeds based on the major species (reactants and products) agree to within a small fraction of a percent. At this point the solution is very close to steady state. The values of the flame speed calculated from the minor species (radicals) are very sensitive, and require more accuracy (both spatial and temporal) in order to achieve the same agreement.

In general, it is easier to obtain an accurate solution for a fast flame than for a slow flame. The slow flame requires many more integration steps before all the oscillations die out and a steady state solution is achieved.

To obtain an accurate solution, both the spatial and temporal accuracy must be sufficient. If the temporal accuracy is low, the calculated flame speeds can oscillate around the correct value. If the spatial accuracy is low, the problem may come very close to convergence, but to the wrong value, since the communication between the collocation points is inadequate. As a consequence of the spatial or temporal accuracy becoming too small, the integration may break down completely.

Spatial accuracy is more important in the flame front, where we have steep gradients. Hence, our algorithms for choosing breakpoints concentrates them in the region to be occupied by the flame front.

The need for spatial and temporal accuracy is connected. Suppose the number of breakpoints is increased, but the temporal error tolerance ϵ is not decreased. The calculated solutions will oscillate, and eventually the solution will break up. On the other hand, suppose ϵ is decreased but the breakpoint sequence is not changed. We can still obtain an

¹⁴J.M. Heimerl and T.P. Coffee, "The Detailed Modeling of Premixed, Laminar Steady-State Flames. I. Ozone", Combustion and Flame, Volume 39, pp. 301-315, 1980.

answer, but the integration will take longer, and the accuracy of the solution is not increased. At this stage, the appropriate value of ϵ for a given breakpoint sequence is a matter of trial and error.

An additional problem can occur with very slow flames. If the interval of integration is chosen too small, there may be a noticeable gradient in the temperature profile at the cold boundary. Since the gas is entering through the cold boundary quite slowly, this can result in a substantial heat loss through the boundary. This heat loss will slow down the flame. The integration will converge, but the computed flame speed will be too low. This effect can occur for any flame. However, for most flames, a noticeable heat loss is due to a fairly steep temperature gradient at the boundary. It is then obvious that the interval of integration must be increased. But for a slow flame (less than 20 cm/sec) a noticeable error in the flame speed can result (5 to 10 percent), even when the gradient at the cold boundary appears negligible.

Most of the cases we have run have been with NINT = 12, KORD = 4, and NCC = 2 (13 breakpoints, 26 collocation points). We have let $\epsilon = 10^{-3}$, and SREC = 10^{-6} . The number of central intervals NCN was 4, and the ratio FC between the longest and shortest intervals was between 4 and 8.

The length of the interval of integration L, as well as the optimum value for FC, is a matter of trial and error. However, a reasonable set of breakpoints can be found fairly easily by experimenting with the above scheme, using the constant transport subroutine. The code can then be run with a more realistic transport algorithm, using the restart option.

To make sure that convergence was obtained, the code was also run with NINT = 16 and $\epsilon = 3 \times 10^{-4}$ for a number of cases. Only negligible differences occurred.

The program was run on the BRL CYBER 76. For hydrogen-oxygen flames, the run time varied from about 30 seconds with the simplest transport subroutine to 5 minutes or more for the most complicated case.

An earlier version of this code was applied to the solution of an unbounded ozone flame¹⁴. The successive calculation procedure was not implemented at that time. However, since the number of species is so small, the savings in reducing the size of the Jacobian is not very important.

At that time we reported using NINT = 57 to NINT = 70, KORD = 6 and NCC = 5. The number of collocation points was then between 62 and 75. We have now been able to solve this case with equal accuracy using NINT = 12, KORD = 4, and NCC = 2. Partly this is due to choosing NCC = 2. The program then generates a special choice of collocation points (Gauss - Legendre quadrature points in each subinterval) which gives increased spatial accuracy. More important is the improved algorithm for choosing breakpoints. By using a little more care, a much smaller set of breakpoints can still reproduce the flame with sufficient accuracy. So the careful choice of breakpoints is probably the most important consideration in running the code efficiently.

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APPENDIX A

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APPENDIX A

A listing of the computer code follows. The subroutines MAIN, F, UINIT, FLSP, BNDRY, BKPT, and RT are written for laminar flame problems. The subroutine F given is for the constant transport assumption. The subroutine RT given is for a H₂-O₂-N₂ flame. There are nine chemical species and thirty reactions.

The rest of the subroutines are from PDECOL. They have been modified as discussed in the text.

After the listing, the job stream and output is given for a typical hydrogen-oxygen flame. The initial unburned gas is 50% H₂ and 50% air, where air is 21% O₂ and 79% N₂. The nitrogen is considered to be a diluent, and does not react. The required starting information is attached to TAPE11. The main program and the subroutine RT are attached in a compiled form. The integration begins from the initial profiles (37). At the final output time, the flame speed and the temperature and species profiles are given, as well as the corresponding x values. The restart file is written on TAPE 2, and the output file on TAPE 9. Both files are catalogued for possible future use.

** 03/10/90 SCOUNT 2.1.5 A H L VEH 004 *** 04/12/90 H0106
 VMS SERVICES R19/ 4/MF FLS=377K FLL=1750K MAX=300K MAX=1.5LK MAX=1.103H
 MM.MM.SS CPU SECND UPDGIN
 11.06.56 MFA. 42 HHL NOS/HF 1.3 L499 VFM 004 03/11/90
 11.07.03 00000.003 MFZ. -TEHY. STMFZ. JU.R. PHYL PLFLSC. CY=5 FOR HHL REPORT.
 11.07.03 00000.003 JOR. -ACCOUNT.PU...
 11.07.04 00000.026 JOH. -ATTACH.OOLDPL1.PLFLSC1.JU=SHE.JMH.
 11.07.05 00000.029 MFZ. MFZ. = CYCLE 5 ATTACH.H FROM SH=SYSTEM
 11.07.05 00000.030 LOD. -UPDATE (P=OLIPL1.U,L=1)
 11.07.42 00000.045 USR. LOD.
 11.07.44 00000.171 USR. READING INPUT
 11.07.44 00000.172 LOD. UPDATE COMPLETE FFL
 11.07.53 00001.990 USR. -FTN(H=COMPILE.SL,R=0,T)
 11.08.54 00001.995 JOH. 1.714 CP SECND COMPILATION TIME
 11.08.54 00001.999 MFZ. -ATTACH.OOLDPL2.PLH202HTW30ND.ID=SHE.JMH.
 11.08.54 00001.999 LOR. PFZ. = CYCLE 1 ATTACH.H FROM SH=SYSTEM
 11.09.01 00001.914 USR. -UPDATE (P=OLIPL3.U,L=1)
 11.09.01 00001.941 USR. READING INPUT
 11.10.01 00001.942 LOU. UPDATE COMPLETE FFL
 11.09.56 00002.416 USP. -FTN(H=COMPILE.SL,R=0,T)
 11.09.56 00002.422 JOH. 1.715 CP SECND COMPILATION TIME
 11.09.57 00002.425 MFZ. -ATTACH.OOLDPL2.PLFLSC2.ID=SHE.JMH.
 11.09.57 00002.426 LOD. PFZ. = CYCLE 1 ATTACH.H FROM SH=SYSTEM
 11.10.11 00002.440 USR. -UPDATE (P=OLIPL2.U,L=1)
 11.10.35 00002.901 USR. READING INPUT
 11.10.35 00002.902 LOD. UPDATE COMPLETE FFL
 11.20.41 00006.645 USR. -FTN(H=COMPILE.SL,R=0,T)
 11.20.41 00006.647 MFZ. 3.739 CP SECND COMPILATION TIME
 11.20.41 00006.647 MFZ. JM166 = MAXIMUM USER SCM 4700DH WORDS
 11.20.41 00006.647 MFZ. JM167 = MAXIMUM USER LCM 7000H WORDS
 11.20.41 00006.647 MFZ. JM170 = MAXIMUM JS+IO LCM 61H HUFFRS
 11.20.41 00006.648 MFZ. JM770 = MAXIMUM ACTIVE FILES 5
 11.20.41 00006.648 MFZ. RM771 = OPEN/CLOSE CALLS 74
 11.20.41 00006.648 MFZ. RM772 = DATA TRANSFER CALLS 37+754
 11.20.41 00006.648 MFZ. RM773 = CONTROL/POSITIONING CALLS 245
 11.20.41 00006.648 MFZ. HM774 = RM DATA TRANSFER CALLS 3+34H
 11.20.41 00006.648 MFZ. HM775 = BM CURSOR/POSITIONING CALLS 342
 11.20.41 00006.649 MFZ. HM776 = QUEUE MANAGER CALLS 746
 11.20.41 00006.649 MFZ. HM777 = RECALL CALLS 574
 11.20.41 00006.649 MFZ. SCM 133.251 KWS
 11.20.41 00006.649 MFZ. LCM 112.623 KWS
 11.20.41 00006.649 MFZ. I/O 0.442 HW
 11.20.41 00006.650 MFZ. RMS 0.117c MWS
 11.20.41 00006.650 MFZ. USEP 3.4201 SEC
 11.20.41 00006.650 MFZ. JOH 6.65r SEC
 11.20.41 00006.650 MFZ. OI0 1.610.496/ KW
 11.20.41 00006.650 MFZ. SS H.15m STC
 11.20.41 00006.651 MFZ. COST ESTIMATE \$1.024
 11.20.41 00006.651 MFZ. SC050 = 00006U SC/LC SWAP,

```

PROGRAM MAIN(INPUT,OUTPUT,TAPL3=INPUT,TAPL3=OUTPUT,TAPL1,
* TAP2,TAP4,TAP9,TAP10,TAP11)
C ****
C THIS VERSION USES SUCCESSIVE CALCULATION.
C EACH PDE IS INTEGRATED IN TIME INDEPENDENTLY.
C ****
COMMON/ENDPT/PHO,PHB
COMMON/TAP/PRESS,PSK,NPDEM
COMMON/GEAR0/DTSEU,NQ,NSTEPS,NF,NJ
COMMON/GEAR1/DU45,I,UKOND,1UUM(1-)
COMMON/OPTION/INDAUS,MAXUE
COMMON/IUNIT/LOUT
COMMON/GEARSS/HMA
COMMON/TACHSM/SMALL
COMMON/START/INSTANT,NBURN,NTRAN,KORU,NSKIP
COMMON/TARAH/ASPRSP,TPN,PHN,MII,TMSPH,TMSPH,TMSPH,TMSTP,TPENT
COMMON/TARFM/FM
COMMON/TARUM,LB1NP(1)
COMMON/TARHK/PHMKPT(31)
COMMON/TABLK/LB12(1)
DIMENSION UPH(20),UUF(20)
COMMON/TACHT/RU,CPMA,HU(20),R2D(20),R2DM(20)
COMMON/TANV,CPIV,K2UF(20)
COMMON/TABMF/RHUV(1000),DRHUV(1000)
COMMON/TARV/PHVAL(401),UN(401),U(20,401),UC(20),UR(20),W(20)
COMMON/MAIN/SCTH(25),WORK(12000),IWORK(1750)
COMMON/START/IN1,Iw2,Iw3,Iw4,Iw5,Iw6,Iw7,Iw8,Iw9,Iw10,
* Iw11,Iw12,Iw13,Iw14,Iw15,Iw16,Iw17,Iw18
COMMON/TARCH/D24
DATA LOUT,NOGAUS,MAXUE,UROUNU/3,0,5,7,1E-15/
DATA NDIM1,NDIM2/20,401/
DATA IWORK(11),IWURK(2)/12000,750/
C ****
C THE FOLLOWING INPUT DATA MUST BE SPECIFIED.
C DIMENSIONING IN MAIN,F,UINIT,FLSP,OR,IHNURY,WRIT,AND INIT.
C SUBROUTINE PT MUST HE WRITTEN.
C IF NTRAN = 2, SUBROUTINE F MUST HE WRITTEN.
C ****
READ(5,402)LB1NP
402 FORMAT(1A10)
        WRITE(3,450)LB1NP
450 FORMAT(1/7A10//)
        READ(5,430)PRESS
        WRITE(3,405)MPDE
405 FORMAT(10X,GMPDE=14/)
        WRITE(3,455)
455 FORMAT(1/26X,1Hw,11X,2HJC,10X,21H,10X,24H,10X,3H,20//)
        DO 425 K=1,NPDF
        READ(11,410)H(K),UC(K),UM(1),MU(K),MC(K)
410 FORMAT(10,1P5E14,H)

```


PROGRAM MAIN 76776 OPT=1 ROUND=---*/ TRACE FIN 4.0.H+49H 04/15/80 11.07.44 PAGE 3

```

115      MF=22
        PSKPRESS/R2.05
        TMSPH=TMSPHN
        TMSPH2=TMSPHPN
        TMSTP=TMVTPN
        UZM=10.0/FINAL
        C  NORMALIZE DIFFUSION AND THERMAL CONDUCTIVITY TERMS.
        DO 9 K=1,NPDE
9       H2D(K)=R2D(K)/PHN
        C  ALL H2D = RH*RH*UIM/PHN
        C  H2D IS THE VALUE AT THE LEFT BOUNDARY.
        C  K2D IS THE VALUE AT THE RIGHT END IF AN INTERVAL USED FOR
        C  COMPUTING THE FLAME SPEED.
        C  FOR NTHAN = 1, THEY ARE THE SAME.
        C  NCPTS=(KORD-NCC)*NINT+NCC
110      DO 602 K=1,NPDE
602      H2D(K)=R2D(K)
        C  FIND THE APPROPRIATE NORMALIZED VALUES FOR USE IN F (INTRAN=1).
        CPINV=TMSTP/CPMAX
        DO 605 K=1,NPDE
605      R2D(K)=R2D(K)*TMSPH
        RL=RHL*TMSPH2/CPMAX
        NPUEM=NPUDE-1
        NPUEP=NPUDE+1
        NCPTS=(KORD-NCC)*NINT+NCC
115      YSSM=0.0
        YN=1.0
        DO 5 K=1,NPDEM
5       YN=YN-UC(K)
        YSSM=YSSM+UC(K)/W(K)
        YSSM=YSSM+YN/W(NPDE)
        T=UC(NPDE)*TPN
        HHO=PSK/(T*YSSM)
        WRITE(13,B)RHO
        FORMAT(15X,17HINITIAL DENSITY =,1PE12.4/)
        NPTS=NINT+1
        T=FINAL*1.0E-10
        TOUT=0.0
        C  HMX IS THE LARGEST TIME STEP ALLOWED IN THE INTEGRATION.
        C  INDEX=1
        INDEX=1
        CALL HKPT(NINT,NGEN,NVPTS)
        C  DEFINE THE CENTER OF THE FLAME.
        KCEN=NPDE
        VCE=N*(KCEN)+0.0*UH(KCEN)-UC(KCEN))
        WHITE(3.573)*KCEN*VCTH
        S73      FORMAT(10X,6HKCEN =,1HE12.4/)
        C  INITIAL VALUE FOR THE SPEED OF THE ORIGIN (MASS FLOW).
        IF(INSTART.EQ.2)READ(1,255)SPEED1
        IF(INSTART.EQ.2)WHITE(3.552)*SPEED1
        IF(INSTART.EQ.1)SPEED0=0.0
        ASPECFIO
        IF(INKNUM.EQ.0.0)GO TO 550
        FM=FMO*FLISP
        WRITE(3,67)FM,FLISP
        SPECIO=FMO*TMSPH
        SPN1=SPEENO
116      MAIN
117      MAIN
118      MAIN
119      MAIN
120      MAIN
121      MAIN
122      MAIN
123      MAIN
124      MAIN
125      MAIN
126      MAIN
127      MAIN
128      MAIN
129      MAIN
130      MAIN
131      MAIN
132      MAIN
133      MAIN
134      MAIN
135      MAIN
136      MAIN
137      MAIN
138      MAIN
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142      MAIN
143      MAIN
144      MAIN
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148      MAIN
149      MAIN
150      MAIN
151      MAIN
152      MAIN
153      MAIN
154      MAIN
155      MAIN
156      MAIN
157      MAIN
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159      MAIN
160      MAIN
161      MAIN
162      MAIN
163      MAIN
164      MAIN
165      MAIN
166      MAIN
167      MAIN
168      MAIN
169      MAIN
170      MAIN
171      MAIN
172      MAIN

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```

PROGRAM MAIN      76/76   OPT=1  KOUND=--*/  FACE          FTN 6.0.H+4.9H
04/15/80  11.07.44  PAGE 4

ASV=SPEEDO
HSP=0.0
CONTINUE
175  FORMAT(1$X,BHSPEL0 =,1PE12.4/)
      175  FORMAT(1$X,BHSPEL0 =,1PE12.4/)
      175  FORMAT(1$X,BHSPEL0 =,1PE12.4/)
      175  FORMAT(1$X,BHSPEL0 =,1PE12.4/)

      175  F1=SECOND(CP)
      175  C  CALL INTEGRATOR AND WRITE OUTPUT.
      175  20  CALL PECOL(I0,TOUT,U,PHKPT,L,*,NINT,KURU,NCC,NPUF,MF)
      175  * INDEX,WORK,INWK,SPEC)
      175  IF (INDEX.NE.0) GO TO 70
      175  GT=SECOND(CP)
      175  HT=(G1-F1
      175  IF (HT.GT.1.0) TFINAL=TOUT
      175  WRITE(3,30) TOUT,DTOUT,NSTEPS
      175  FORMAT(1/10X,3H1 =,1H12.4,4X,4,..,1 =,1H12.4,4X,
      175  * 13H TOTAL STEPS =,1M1)
      175  WRITE(3,3?1NF*NJ
      175  32  FORMAT(1/10X,4HMF =,1M,5X,4HNJ =,1B/)
      175  WRITE(3,45) HT
      175  FORMAT(1/10X,10HNU TIME =,1PE12.4/
      175  45  CALL VALUES(PHVAL,U,SCATCH,NDIM1,*,NIM2,NVPTS,0,WORK)
      175  DU 25 N=1,NVPTS
      175  UN(K)=1.0
      175  DU 25 J=1,NODEM
      175  UN(K)=UN(K)-U(J,K)
      175  IF (TOUT.LT.TPRINT) GO TO 65
      175  DU 35 K=1,NODEM
      175  WRITE(3,371)B(K)
      175  FORMAT(1/10X,A10/)
      175  WRITE(3,39) (UN(I),I=1,NVPTS,NSKIP)
      175  FORMAT(1$X,BHSPEL0 =,1PE12.4)
      175  CONTINUE
      175  WRITE(3,37)LA(NUDE)
      175  WRITE(3,39) (UN(I),I=1,NVPTS,NSKIP)
      175  WRITE(3,371)B(NSKIP)
      175  WRITE(3,39) (UN(I),I=1,NVPTS,NSKIP)
      175  CONTINUE
      175  CALL FLSP(NSP,UPH,NUF)
      175  * PHOKEN,FS,UPH,NUF)
      175  FORMAT(1$X,BHSPEL0 =,1PE12.4,6,4HUS =,1PE12.4/)

      175  65  IF (TOUT.GE.TPRINT) TFINAL=TPRINT/5.0
      175  IF (NHURN.EQ.0) GO TO 100
      175  DU 610 J=1,NODEM
      175  UDF(IJ)=U(IJ,1)-UPH(IJ)*R2D(IJ)/FM
      175  UDF(NSKIP)=1.0
      175  DU 612 J=1,NODEM
      175  UDF(NSP)=UDF(NSP)-UDF(IJ)
      175  WRITE(3,615) (UDF(IJ),I=1,NPDE)
      175  615  FORMAT(1/2*29HSP,C15 MASS FLUX + FAC15 =,1H12.4/1)
      175  C  FINU THE FORMULA FOR ADJUSTING THE SH15N SPFLU AND CENTFLU
      175  C  TH FLAME.
      175  100  DU 110 K=1,NVPTS
      175  K=P+1
      175  UM=AMAX1(U(KCEN,K),U(KCEN,M))
      175  UMIN=AMIN1(U(KCEN,K),U(KCEN,M))
      175  IF (UMAX.GT.UCEN.AND.UMIN.LT.UCEN)KS=K
      175  IF (UMAX.GT.UCEN.AND.UMIN.LT.UCEN)KS=K
      175  110

```

PROGRAM MAIN 76/76 OPT=1 ROUND=0.01 ACT FIN 4.0449H 04/15/80 11.07.44 PAGE 5

```

115 K=KS+1
  PC=(VCE-N*(KCE-N))/((VCE-N)*(KPC)-N*(KCT-N))
  PHNEW=PHVAL(KS)*PC*(PHVAL(KP))-N*VAL(KS)
  IF INHURN EQ .1) GO TO 560
  IF ITOUT GT 0.0) GO TO 150
  FM=HNO*FSP
  IF INSTART EQ .1) SPEEDU=-FM*TMSPH
  WRITE(13*447)FM,FSP
  FORMAT(1/5X,4HNO =,1PF12.4,1)
  DS=0.0
  IF ISPEEDU GT 0.0) SPEEDU=0.0
  WRITE(13*443) SPEEDU,US
  TINC=FINAL/100.0
  TOLD=TOUT
  TUUT=TINC
  UPH=PHNEW-PHCT
  UT=TOUT-TOLD
  PHOLD=PHNEW-SPEEDU*U1
  HSP=0.0
  ASP=ASPEEDO
  SPEEDN=ASP*BSP*UT
  SPN1=SPEEDN
  GO 10 20
  SPEED0=SPEEDN
  SP01=SPN1
  SPH1=(PHNEW-PHOLD)/(TUUT-TOLD)
  FM=SPN1/TMSPH
  FSP=FM/RHO
  WRITE(13*47)FM,FSP,PHHTW
  FORMAT(1/5X,4HNO =,1PF12.4,6X,6HFLSP =,1PF12.4,6X,1
  * 7PHNEW =,1PF12.4,1)
  DS=SPN1-SP01
  TOLD=TOUT
  C ESTIMATE CHANGE IN FLAME SPEED PER UNIT TIME.
  USG=AH(SUS)/TINC
  C CHOOSE A NEW TINC SO THAT THE ESTIMATED DRIFT OF THE FLAME
  C FROM THE CENTER IS VERY SMALL.
  UPHMAX=(PH3-PH2)/50.0
  TINC=SIGN(DPHMAX/USG)
  TINC=A MAX(TINC,TFINAL/100.0)
  TINC=A MIN(TINC,TFINAL/4.0)
  TUUT=TOUT+TINC
  IF (TUUT GT TFINAL) TUUT=FINAL
  IF (TOLD EQ .0) FINAL=0.0
  UT=TOUT-TOLD
  IPH=PHNEW-PHCT
  SPN2=2.0*SPN1-SPEEDU*2.0*DPM/UT
  SPEDUN=0.5*(SPN1+SPN2)
  C IF UPH IS SMALL ENOUGH, DON'T HURT TO RECENTR FLAME.
  IF (ARCSUPH LT 0.01*(IPHS-PHD)) SPN2=SPN1
  C FLAME SPEED CANNOT BE NEGATIVE.
  IF (ISPEEDU GT 0.0) SPEEDU=0.0
  *SP=(SPEED0-SPEEDN)/(TUUT-TOLD)
  ASP=SPEEDU-TOLD,SP
  US=SPEEDUN-SPEEDN
  WRITE(13*43)SPN1,SPN2,SPEEDUN
  FORMAT(1/5X,6HSPN1 =,1PF12.4,6X,1,SPN2 =,1PF12.4,6X,1
  * 7PHNEW =,1PF12.4,6X,1
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PROGRAM MAIN 76/76 OPT=1 NOUND=--*/ 11.ACE FTN 4.0.H+44H 04/15/H0 11.07.44 PAGE 6

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* BHSPEEDN = 1PF12.4+6X4HDS = 1E12.4/
* PHOLD=PHINF-W-DT*(ASP+BSR*(ITOUT+1ULD)*0.5)
  GO TO 570
  560 TOUT=TOUT+TFINAL/N+0
  570 CONTINUE
    IF ITOUT.L.E. TFINAL() GO TO 20
  <01 CONTINUE
C CREATE DATA FILE FOR HSTART
  WRITE(13*220) SPHL
  WRITE(12*220) SPNI
  WRITE(13*220) PHNEW
  WRITE(12*220) PHNEW
  WRITE(13*220) FMTI/10X,1AHCOLLULATION POINT,/)
  200 FORMAT(I10)
  WRITE(12*210) NCPTS
  WRITE(13*210) NCPTS
  FORMAT(1A)
  210 IEND=IW3+NCPTS-1
  WRITE(12*220) (WORK(L)*L=IW3, IENU)
  WRITE(13*220) (WORK(L)*L=I3J, JENU)
  CALL VALUES(WORK(IW3), U, SC1CH, N, IM1), NU1M2, NCPTS, 0, WORK)
  DO 250 K=1,NPDE
  WRITE(12*220) (UK*L)*L=1, NCPTS)
  WRITE(13*220) (UK*L)*L=1, NCPTS)
  CONTINUE
  250 WRITE(12*220) (R2D(L)*L=1, NPDE)
  WRITE(13*220) (R2D(L)*L=1, NPDE)
  FORMAT(1PF14.6)
  220 CONTINUE
  500 WRITE(13*80) INDEX
  STOP
  315 FORMATT/10X,7MINDEX = , I3)
  END
  60

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MAIN

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SUBROUTINE F 7676 OPT=1 MOUND=-*/ TRACE FTN 6.8+49H 04/15/60 11.07.44 PA,F 7
 1 SUBROUTINE F (TIME, UPH, UPHE, VAL, INPUT, KPUT, IC, KSKT, SKW)
 2 DIMENSION U(INPUT), UPH(INPUT), UPHE(INPUT)
 3 COMMON/TABAH/ASP, HSP, TPH, PHM, TM, TMPH, TMPH2, TMSPH, TMSPH2, TPENT
 4 COMMON/TAMP/PRESS, SW, NPDEM
 5
 6 DIMENSION H(20), R(20)
 7 COMMON/TARC/RL, CPMX, HO(20), H2U(0), H2UH(20)
 8 COMMON/TBNT/CPINV, H2UF(20)
 9 COMMON/TARRY/T, WH, Y1, Y2, Y3, Y4, Y5, Y6, Y7, YH, Y9, Y10
 10
 C *****
 C AT EACH CALL THE TIME KATT OF CHAN,F FOR ONE PDE IS RETURNED IN FVAL.
 C *****
 11
 12 VN=1,0
 13 UU 10 K=1,INPUT
 14 YN=YN-U(K)
 15 CALL RTI(N, K, NPDE, UPHE, IC, KSKW)
 16 SP=ASP*HS*ETIME
 17 IF (KPUF .EQ. NPDE) GO TO 50
 18 FVAL=SP*UPH(KPDE)*H2UF(KPDE)*UPH1(KPDE)*TMNHR(KPUF)
 19 HT TURN
 20
 21 UT=IPN*UPH(KPDE)-TPENT
 22 UU 20 K=1,NPDE
 23 H(K)=HO(K)+CPMX*UT
 24 HTEMP=0.0
 25 UU 25 K=1,NPDE
 26 HT=TEMP*UT*(K)/H(K)
 27 FVAL=SP*UPH(NPDE)*HL*UPH2(INPUT)*HTEMP*CPINV
 28 KETURN
 29 ENJ

```

SUBROUTINE UNIT      7676   OPT=1 4.0UNITS=0-#/- 1.ACT          FIN 4.0R+44H
          04/15/60  11.07.64   PAGE  R

1      SUBROUTINE UNIT(PH1,PH2,PH3,PH4,PH5)
2      COMMON/LAIN/PH2,PH3,PH4,PH5
3      COMMON/NUPT/PHU,PHS
4      COMMON/THA/ASP,HSF,TPH,PHN,IM,TMSPH,TMSPH2,TMSPH3,TPENT
5      COMMON/STANT/NS,KHUKKINTHAN,KUH,NSKIP
6      COMMON/TARV/PHVAL(401),PHS(401),UP(20,401),UC(20),UH(20),W(20)
7      PIS2=PI / 2.0.
8      DATA PIS2/1.5707963/
9      *****
10     C DETERMINE THE INITIAL STARTING PROFILF.
11     C X WILL BE A COLLOCATION POINT.
12     C USES THE CORRESPONDING FUNCTIO., VALUES.
13     C *****
14     IF (NS.GT.1)GO TO 50
15     C DETERMINE AN INITIAL GUESSED PROFILF. FROM VALUES AT THE
16     C BURNED AND UNBURNED ENDS.
17     IF (PH1.GT.PH2)GO TO 5
18     U=UC(KPDE)
19     RT TURN
20     IF (PH1.GT.PH3)GO TO 10
21     PHU=(PH1-PH2)/(PH3-PH2)
22     PHM=PIS2*(PHD**2)
23     UD=UH(KPDE)-UC(KPDE)
24     U=UC(KPDE)+UD*((SIN(PHM))***2)
25     RT TURN
26     U=UH(KPDE)
27     RT TURN
28     IF (NS.GT.2)GO TO 100
29     C HEAD THE COLLOCATION VALUES AND THE CORRESPONDING FUNCTION
30     C VALUES FROM A PREVIOUS RUN.
31     HEAD(1,65)PHCTO
32     HEAD(1,65)NPS
33     FORMAT(1B)
34     NM=NPS-1
35     HEAD(1,65)(PHS(L),L=1,NPS)
36     DO 60 K=1,NPDE
37     HEAD(1,65)(UP(K,L),L=1,NPS)
38     C WE DO NOT WANT ANY NEGATIVE CONCENTRATIONS.
39     DO 58 L=2,NM
40     IF (UP(KL).GE.0.0)GO TO 5H
41     LM=L-1
42     UP(K,L)=0.5*(UP(K,LH)+UP(K,LP))
43     CONTINUE
44     FORMAT(1B)
45     IF (NHURN.FG.1)GO TO 75
46     C CNT IN THE FLAME.
47     DO 70 K=1,NPS
48     PHS(K)=PHS(K)+PHCTO
49     FORMAT(1B)
50     CONTINUE
51     IF (NHURN.FG.1)GO TO 75
52     CNT IN
53     PHXL=PHS(2)-PHU
54     PHXR=PHS(NPS)-PHS(NPS)
55     DO 56 K=2,NPS
56     U=UC(KPDE)
57     RT TURN

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SUBROUTINE UNIT 76/76 OPT=1 K(UNI)=-*/ IMAGE
 FIN 4.0.H+498
 04/15/80 11.07.44 PAGE 0

```

C DETERMINE THE NEW VALUES BY INTERPOLATION.
100   K=2
      IF (PH .GT. PH0) GO TO 102
      U=UC (KPDE)
      RETURN
102   IF (PH .GT. PHS(2)) GO TO 104
      P=(PH-PH0)/PHXL
      U=UC (KPDE)+P*(UW (KPUT,2)-UC (KPDU))
      RETURN
104   IF (PH .LT. PHS(NPS)) GO TO 108
      IF (PHAR .NE. 0.0) W=(PH-PHS(NPS))/R-XR
      IF (PHAR .EQ. 0.0) P=0.0
      U=UP (KPDE,NPS)+P*(UW (KPUE)-UP (KPUE,NPS))
      RETURN
108   CONTINUE
110   IF (PH .LE. PHS(K)) GO TO 120
      K=K+1
      GO TO 110
120   KM=K-1
      P=(PHS(K)-PH)/(PHS(K)-PHS(KM))
      U=UP (KPUE,K)-P*(UP (KPUE,K)-UP (KPUE,KM))
      RETURN
END
  
```

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SUBROUTINE FLSP      76/76   OPT=1 ROUND=--> TRACE          FTN 4.H+4.H
                                                               04/15/R0  11.07.44
                                                               PAHF    10

1      SUBROUTINE FLSP (NUPH,TOUT,TPIINT,TFINAL,NVPTS,NCPTS,
*     KHO,KCF,FSP,UH,UH)
2      C ***** DETERMINE THE FLAME SPEED BY NUMERICALLY INTEGRATING THE SPECIES
3      C EQUATIONS, USING THE THREE-ZONAL MODEL.
4      C ****
5      DIMENSION UPINDE(1) UDF(1) NDE(1)
6      COMMON/EINPUP/PLM,PMK
7      COMMON/TAIN/PFL,MPFF,MPFT,TPN,PHN,TR,TPSPH,TASHW,TMSTP,TENT
8      COMMON/TARAH/SUPHSP,TPN,PHN,TR,TPSPH,TASHW,TMSTP,TENT
9      COMMON/TABP/PRESS,PSK,NPDEM
10     COMMON/TARIM/NDIM,NUIM2
11     COMMON/START/NSTART,NHURN,INTRAN,ORD,NSKIP
12     COMMON/MAIN/SCICH(25),WORK(12000),IWORK(750)
13     COMMON/ISTART/IW1,IWC,IW3,IOWM(1)
14     COMMON/TARCT/RL,CPLM,HO(120),KEDU(120)
15     COMMON/TABV/PVAL(401),UN(401)(120,401),UC(20)(120)
16     COMMON/TARWF/RMUV(1000),DRMV(1000)
17     COMMON/TARSM/SMALL
18     COMMON/TARBL/B(21)
19     DIMENSION RINT(120),KINT1(20),KI(120)
20     DIMENSION ULT(120,3),URT(20,3),FLN(120),UPH2(120)
21     DIMENSION X(401),W(120),D120)
22     DO 5 J=1,NPDE
23     H1(J)=0.0
24     C INTEGRATE FROM THE LEFT BOUNDARY TO VARIOUS POINTS.
25     CALL VALUES(IPLH,ULT,SCICH,NDIM,1,2,WORK)
26     FORMAT(1X,13HKE20 AT LEFT =,1P1L12.4/)
27     DO 10 J=1,NPDE
28     UPH2(J)=ULT(J,2)
29     UPH(1PDE)=0.0
30     UPH2(NPDE)=0.0
31     DO 12 J=1,NPDE
32     UPH(1PDE)=UPH(NPDE)-UPH(J)
33     IF (LAHS(UPH(NPDE)) .LT. SMALL) UPH(1PDE)=SMALL
34     R2D(NPDE)=RADM(1PDE)/UPH(1PDE)
35     WRITE(13,128) (K2D(L),L=1,NPDE)
36     FORMAT(13*126) UPH(L),L=1,NPDE),ULT(1PDE*2)
37     R2D(1PDE)=RADM(1PDE)/UPH(1PDE)
38     WRITE(13,129) (K2D(L),L=1,NPDE)
39     FORMAT(13*126) UPH(L),L=1,NPDE),ULT(1PDE*2)
40     R2D(1PDE)=RADM(1PDE)/UPH(1PDE)
41     WRITE(13,127) (UPH2(L),L=1,NPDE),ULT(1PDE*2)
42     R2D(1PDE)=RADM(1PDE)/UPH(1PDE)
43     WRITE(13,128) (UPH2(L),L=1,NPDE),ULT(1PDE*2)
44     R2D(1PDE)=RADM(1PDE)/UPH(1PDE)
45     NS=32
46     NSPL=NS+1
47     DO 155 NMID=NSPL,NVPTS,NS
48     NMIDU=NMID-NS
49     DO 100 K=NMI00,NMIU
50     CALL RT(U(1,K),UN(K),N+NPDE+1,1,1)
51     DO 20 J=1,NPDE
52     HINT(J)=H(J)
53     CONTINUE
54     IF (K.EQ.NMIU) GO TO 100
55     IF (K=1)
56     UPH=PVAL(K)-PVAL(K-1)
57     U(J)=U(J)+0.5*(W1(J)-(J)*RT(U(1,J))-UPH
58

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      SUBROUTINE FLSUP      76/76   OPT=1  H(UNI)=--*/ 1-ACE          FIN 4.0.H.044H
      115      FORMAT(10X,7HX 1' CM/1)          04/15/40  11.07.44
      116      WRITE(3,202) X(L),L=1,NVPTS,NK1W)    FLSUP
      117      FORMAT(IP10E12.4)                 FLSUP
      118      CONTINUE                           FLSUP
      119      C COMPUTE THE FLAME THICKNESS.
      120      TMX=MAX1(U(NPDE,NVPTS)*U(INPUT+1))
      121      TMN=MIN1(U(NPDE,NVPTS)*U(INPUT+1))
      122      UT=0.1*(TMX-TMN)
      123      TH=TMX-UT
      124      TL=TMN+UT
      125      DO 310 K=1,NVPTS
      126      KP=K
      127      IF (TL.GT.U(NPDE,K).AND.TL.LT.U(NPDE,KP)) KL=K
      128      IF (TM.GT.U(NPDE,K).AND.TM.LT.U(NPDE,KP)) KM=K
      129      CONTINUE
      130      KL=KP+1
      131      KM=KP+1
      132      PL=U(NPDE,KL)-TL/(U(NPDE,KL)-U(1,KLP))
      133      PH= U(NPDE,KM)-TM/(U(NPDE,KM)-U(1,KMP))
      134      PH>PHVAL(KL)*PL*(PHVAL(KLP)-PHVAL(KL))
      135      PH<PHVAL(KM)*PM*(PHVAL(KMP)-PHVAL(KM))
      136      XU=X(KL)*PL*(X(KLP)-X(KL))
      137      XM=X(KM)*PM*(X(KMP)-X(KM))
      138      FT=H(S(XL-XH)
      139      WRITE(3,314) PHIL,PHIM
      140      PHEN=0.5*(PHL+PHM)
      141      PHFL=AMIN1(PHL,PHM)
      142      PHFM=AMAX1(PHL,PHM)
      143      FORMAT(10X,23MF,FLAME FRONT FROM MH1=,1PE12.4,2X,
      144      * 0.01PE12.4,1)
      145      WRITE(3,315) XLM,XMH
      146      FORMAT(10X,20HFLAME FRONT FROM X =,1PE12.4,2X,
      147      * 0.01PE12.4,1)
      148      WRITE(3,330) FTH
      149      FORMAT(10X,17HFLAME THICKNESS =,1PE12.4,2X,2HCM/1)
      150      IF (ITOUT.LT.TFINAL) THEN
      151      C CMFILE OUTPUT FILE.
      152      WRITE(9,1208) NPDE,NVPTS
      153      WRITE(3,1208) NPDE,NVPTS
      154      FORMAT(5I4)
      155      OU 1215 K=1,NPDE
      156      WRITE(9,1212) LK(K),M(K),MO(K),MK(K)
      157      WRITE(9,1212) LK(K),M(K),MO(K),MK(K)
      158      FORMAT(A10,1P3E16,K)
      159      CONTINUE
      160      LM=1000
      161      WRITE(9,1212) LM
      162      WRITE(3,1212) LM
      163      WRITE(9,1210) OUT
      164      WRITE(9,1210) OUT
      165      WRITE(9,1210) ASR,MSP,TPN,PNH,1M,
      166      WRITE(3,1210) ASR,MSP,TPN,PNH,1M,
      167      WRITE(9,1210) PNS,PS,PS
      168      WRITE(9,1210) CPML,TPNT
      169      WRITE(3,1210) CPML,TPNT
      170      WRITE(9,1210) CPML,TPNT
      171      FORMAT(1PF16.4)
      172

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14

DATE

04/15/96 11:07:44

Subroutine BNDY 7K7K OPT=1 HURNH=000/ 1..ACT

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1      SUBROUTINE BNDY (TIT, PH(1), UPH(1), UH(1), NPDE)
2      DIMENSION UPD(1), UPH(1), UH(1), NPDE
3      DATA (NPDE, PH(1)) /0, 0/
4      COMMON/ENDP/PH(1), PHM
5      COMMON/TAPR/PRESS, PS(1), NPDE
6      C   FOR BOUNDARY CONDITIONS. SEE HURNHRLDEH. C JMTSS.
7      AND BIRD. PP. 762. Pg.3.
8      COMMON/TAPC/RC(1), MA(1), HO(120), R20(120), R20M(120)
9      COMMON/TASMF/RHUV(1000), DHUV(1000)
10     COMMON/TASHM/SMALL
11     COMMON/TARFM/FM
12     COMMON/START/NSTANT, NHURN, NTRAN, KOKU
13     COMMON/TARCH/DZM
14     COMMON/TAHV/PAVAL(401), UN(401), ((20+401)), UC(201), UH(201), *(201)
15     DO 5 I=1,NPDE
16     DZDT(I)=0.0
17     DHOU(I)=0.0
18     DHUPH(I)=0.0
19     IF PH.GT.PHL GO TO 50
20     IF (NHURN.EQ.1) GO TO 7
21     DO 10 J=1,NPDE
22     DHOU(J)=1.0
23     RETURN
24     BNDY
25     C HURNH STABILIZED FLAME. WE NEED TO FIND THE MASS FLUX
26     C FRACTIONS AT THE COLD BOUNDARY.
27     C IF (NTRAN.EQ.1) GO TO 12
28     C FIND R20 AT THE LEFT BOUNDARY.
29     DO 11 J=1,NPDE
30     IF (ABS(SUPH(J)).LT.SMALL) UPH(J)=SMALL
31     R20(J)=RHUV(J)/UPH(J)
32     CONTINUE
33     DHOU(NPDE)=1.0
34     DO 12 J=1,NPDE
35     DHOU(J)=1.0
36     DZDUPH(J)=-R20(J)/FM
37     C IF THE MASS FLUX FRACTION EP DOES NOT HAVE THE CORRECT VALUE.
38     C WE USE DZDT TO MOVE IT TOWARDS Y COLD.
39     C UZM IS CHOSEN SO THAT, IF R20 DOES NOT CHANGE, EP WOULD EQUAL
40     C Y COLD AFTER A TIME INTERVAL OF 10.0.
41     DO 25 J=1,NPDE
42     EP=U(J)-2*(J*UPH(J))/FM
43     DZDT(J)=(UC(J)-EP)*UZM
44     RETURN
45     DO 30 J=1,NPDE
46     DHUPH(J)=1.0
47     NHURN
48     END

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1      SUBROUTINE HKPT 76/76 0PR=1 NU(NU)=0--> 1-ACE   FTN 4.0.H.44H
2      ?          HKPT ? 3
3      COMMON/TAHK/PHKPT(3)
4      COMMON/TAHV/PHVAL(401),UN(401)*,(20+401),UC(20),U4(20)*W(20)
5      COMMON/TAHN/PH2,PH3,PHC,PHCPT,PHL
6      COMMON/START/INSTANT,NHURN,NTHAN,KORD,NSKIP
7      COMPUTE THE BREAKPOINTS.
8
9      IF (NHURN,F0.116) TO 650
10     UNHOUNDEU FLAME.
11     PHC=PH0+0.5*(PH5-PH0)
12     PH2=PH0+0.24*(PH5-PH0)
13     P=(PH5-PH0)/(FLUA1(NCH)+2.0*ALP*(ALP**0.1-1.0)/(ALP-1.0))
14     PHHKPT(1)=PH0
15     HKPT 14
16     HKPT 15
17     HKPT 16
18     HKPT 17
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SUBROUTINE HKPT
 76/76  UP=1  HROUND=-*- / 1*ACT
      F7N 4.0.H+4.4H
      04/15/H0 11.07.44
      PAUL   16

C  DEFINE THE EVALUATION POINTS.
      NSP=R
      NVP1S=NSP*NINT+1
      DO 17 K=1,NVP1S
      KV=1+(K-1)*NSP
      PHVAL(KV)=PHVALPT(K)
      NSPP=NSP+1
      DO 19 K=NSP,NVP1S,1
      KM=K-NSP
      UP=(PHVAL(K)-PHVAL(KM))/FLOAT(NVP1S)
      DO 19 J=2,NSP
      KT=KM+J-1
      PHVAL(KT)=PHVAL(KM)+UP*FLOAT(J-1)
      CONTINUE
      NSKIP=NINT/10
      IF (NSKIP.LT.1)NSKIP=1
      WRITE(3,J91)(PHVAL(I),I=1,NVP1S,1,NSKIP)
      FORMAT(1P10E12.4)
      RETURN
      END

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SUBROUTINE RT(U,YNEK,NPUE,KPUE,IU,KS1P)
DIMENSION U(1),R(1)
COMMON/TAP/PRESS,P5K,NPUEM
COMMON/TARH/ASP,HSP,TPN,PHN,1M1,TMSPH,TMSPH2,TMS1P,1P81
COMMON/TARH/T1,T2,HY1,Y2,Y3,Y4,Y5,Y6,Y7,Y8,Y9,Y10
C THE DIMENSION OF RK MUST BE AT LEAST THE NUMBER OF REACTIONS + 1
C THE NUMBER OF COLLOCATION POINTS.
C DIMENSION RK(2000)
10  C HZ-U2 KINETICS.  WANNATZ.  LYUD'S.
      IST= 30*(IC-1)
      T=UNPIE)*TPN
      Y 1=U( 1)/ 1.00
      Y 2=U( 2)/ 17.00
      Y 3=U( 3)/ 16.00
      Y 4=U( 4)/ 33.00
      Y 5=U( 5)/ 34.00
      Y 6=U( 6)/ 2.00
      Y 7=U( 7)/ 32.00
      Y 8=U( 8)/ 18.00
      Y 9=YN/ 28.00
      Y10=Y1+Y2+Y3+Y4+Y5+Y6+Y7+Y8+Y9
      Y11=Y6+0.4*Y7+6.0*YH+0.4*Y9
      HH=PSR/(T*Y10)
      HH2=RH*RH
      IF (KS1P.GT.1) GO TO 10
      RK(1)= 2.2000t+13*EXP( -5.5900E+03/1)
      RK(2)= 9.3000t+13*EXP( -1.0250E+04/1)
      RK(3)= 2.2000t+14*EXP( -n.4500E+03/1)
      RK(4)= 1.0000t+13
      RK(5)= 1.0000t+16*( 1.0000E-00)*EXP( -4.4800E+03/1)
      RK(6)= 8.3000t+09*( 1.0000E-00)*EXP( -3.5000E+03/1)
      RK(7)= 6.3000t+12*EXP( -n.5000E+02/1)
      RK(8)= 6.0000t+13*EXP( -n.2400E+02/1)
      RK(9)= 9.0000t+17*( 1.0000E-00)
      RK(10)= 2.2000t+22*( 1.0000E-00)
      RK(11)= 5.0000t+15*EXP( -n.0000E+02/1)
      RK(12)= 2.5000t+14*EXP( -n.5000E+02/1)
      RK(13)= 2.6500t+13*EXP( -n.5000E+02/1)
      RK(14)= 1.5000t+13*EXP( -n.0000E+02/1)
      RK(15)= 5.0000t+13*EXP( -n.0000E+02/1)
      RK(16)= 5.0000t+12*EXP( -n.0000E+02/1)
      RK(17)= 9.0100t+14*EXP( -n.5500E+02/1)
      RK(18)= 3.2000t+14*EXP( -n.5000E+02/1)
      RK(19)= 8.0500t+12*EXP( -n.0000E+02/1)
      RK(20)= 8.0000t+14*EXP( -n.3000E+02/1)
      RK(21)= 7.0000t+13*EXP( -n.0000E+02/1)
      RK(22)= 7.0000t+15*EXP( -n.3000E+02/1)
      RK(23)= 1.2000t+13*EXP( -n.0200E+04/1)
      RK(24)= 7.0000t+11*EXP( -n.44000E+03/1)
      RK(25)= 1.7000t+12*EXP( -n.9000E+01/1)
      RK(26)= 1.2000t+17*EXP( -n.2900E+04/1)
      RK(27)= 1.2000t+17*EXP( -n.2000E+04/1)
      RK(28)= 2.0000t+13*EXP( -n.1200E+04/1)
      RK(29)= 1.2000t+13*EXP( -n.2000E+04/1)
      RK(30)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(31)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(32)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(33)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(34)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(35)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(36)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(37)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(38)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(39)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(40)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(41)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(42)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(43)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(44)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(45)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(46)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(47)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(48)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(49)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(50)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(51)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(52)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(53)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(54)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(55)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(56)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(57)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(58)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(59)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(60)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(61)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(62)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(63)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(64)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(65)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(66)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(67)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(68)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(69)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(70)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(71)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(72)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(73)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(74)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(75)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(76)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(77)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(78)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(79)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(80)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(81)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(82)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(83)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(84)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(85)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(86)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(87)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(88)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(89)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(90)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(91)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(92)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(93)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(94)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(95)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(96)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(97)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(98)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(99)= 1.2000t+13*EXP( -n.1200E+04/1)
      RK(100)= 1.2000t+13*EXP( -n.1200E+04/1)
      CONTINUE

```

```

SUBROUTINE RT    76/76  OPT=1 ROUND=-*/ TRACE

        60      1=RH*RK(1ST*   1)*Y 2*Y 6
        2=RH*RK(1ST*   2)*Y 1*Y 6
        3=RH*RK(1ST*   3)*Y 1*Y 7
        4=RH*RK(1ST*   4)*Y 3*Y 2
        5=RH*RK(1ST*   5)*Y 3*Y 6
        6=RH*RK(1ST*   6)*Y 1*Y 2
        7=RH*RK(1ST*   7)*Y 2*Y 2
        8=RH*RK(1ST*   8)*Y 3*Y 8
        9=RH2*RK(1ST*   9)*Y 1*Y 1*Y 11
        10=RH2*RK(1ST*  10)*Y 1*Y 2*Y 11
        11=RH2*RK(1ST*  11)*Y 1*Y 7*Y 11
        12=RH*RK(1ST*  12)*Y 1*Y 4
        13=RH*RK(1ST*  13)*Y 1*Y 4
        14=RH*RK(1ST*  14)*Y 1*Y 4
        15=RH*RK(1ST*  15)*Y 2*Y 4
        16=RH*RK(1ST*  16)*Y 3*Y 4
        17=RH2*RK(1ST*  17)*Y 2*Y 2*Y 10
        18=RH*RK(1ST*  18)*Y 1*Y 5
        19=RH*RK(1ST*  19)*Y 4*Y 4
        20=RH*RK(1ST*  20)*Y 6*Y 11
        21=RH2*RK(1ST*  21)*Y 3*Y 3*Y 10
        22=RH*RK(1ST*  22)*Y 4*Y 11
        23=RH*RK(1ST*  23)*Y 2*Y 2
        24=RH*RK(1ST*  24)*Y 4*Y 6
        25=RH*RK(1ST*  25)*Y 1*Y 5
        26=RH*RK(1ST*  26)*Y 5*Y 10
        27=RH*RK(1ST*  27)*Y 3*Y 5
        28=RH*RK(1ST*  28)*Y 3*Y 5
        29=RH*RK(1ST*  29)*Y 2*Y 5
        30=RH*RK(1ST*  30)*Y 4*Y 8
R(1)= 1.00*
        1+H 1-R 2-R 3+H 4+H 5-K 6-C+K
        -H 14-R 16+2.*H 20+H 22+R 23+K 24+R
H(2)= 17.00*
        (-H 1+H 2+R 3-H 4+H 5-K 6-C+K
        +H 15+H 16+2.*H 17+K 18-2.*H 21+2.*H
H(3)= 16.00*
        (+H 3-H 4-H 5+H 6+H 7-H H+H 14-
        )+H 4)= 33.00*
        (+H 11-H 12-R 13-K 14-H 15+H 16-C+H
        +H 5)= 34.00*
H(6)= 2.00*
        (-H 1+H 2+R 3-H 4+H 5-K 6-C+K
        +H 7)= 32.00*
        (-H 3+H 4-H 11+H 13+H 15+H 16+H 19+
H(9)= 0.00*
        (+H 1-H 2+R 7-H 8+H 10+H 14+H 15+
MTURN
END

```

FTN 4.0 H 4.4 W
H=H 10+2.0W
H=H 29+P 30)
1 21+R 27+H 2
1 23+H 24+H 2
1 30) P
1 2R)
1 +H 29+H 30)

PA, f	11.09.01
P1	59
P1	60
P1	61
P1	62
P1	63
P1	64
P1	65
P1	66
P1	67
P1	68
P1	69
P1	70
P1	71
P1	72
P1	73
P1	74
P1	75
P1	76
R1	77
R1	78
R1	79
R1	80
R1	81
R1	82
R1	83
R1	84
R1	85
R1	86
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R1	88
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R1	96
R1	97
R1	98
R1	99
R1	100
R1	101
R1	102
R1	103
R1	104
R1	105
R1	106
R1	107
R1	108
R1	109
R1	110
R1	111

CALL NO.	SERIORITY	ROUTINE	DETAILS	DIAGNOSIS OF PROBLEM	FTN 4.R4498	04/15/80	11.09.01	PAGE
14	1	76/76	OPT=1 HOUND=++*/ ACT	ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				3
15	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
15	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
15	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
17	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
17	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
19	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
20	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
91	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
94	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
97	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
100	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
102	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
104	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
105	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				
105	1			ARRAY REFERENCE OUTSIDE DIMENSION HOUND.				

76/76 OPT=1 MOUNU=--*/ 1-MACT FTN 4.0-H049H 04/15/80 11.10.37 PAGE 2

C WHEN H AND Z ARE ARBITRARY VECTOR VALUED FUNCTIONS WITH
 C INPUT COMPONENTS AND U, UX, AND UZ ARE AS ABOVE. THESE BOUNDARY
 C CONDITIONS MUST BE CONSISTENT WITH THE INITIAL CONDITIONS WHICH ARE
 C DESCRIBED NEXT.

C INITIAL CONDITIONS

65 C EACH SOLUTION COMPONENT U(K) IS ASSUMED TO BE A KNOWN (USER
 C PROVIDED) FUNCTION OF X AT THE INITIAL TIME T = T0. THE
 C INITIAL CONDITION FUNCTIONS MUST BE CONSISTENT WITH THE BOUNDARY
 C CONDITIONS ABOVE. I.e. THE INITIAL CONDITION FUNCTIONS MUST
 C SATISFY THE BOUNDARY CONDITIONS FOR T = T0. SEE SUBROUTINE UINIT
 C DESCRIPTION BELOW.

C REQUIRED USER SUPPLIED SUBROUTINES

75 C THE USER IS REQUIRED TO CONSTRUCT THREE SUBPROGRAMS AND A MAIN
 C PROGRAM WHICH DEFINE THE PDE PACKAGE, WHOSE SOLUTION IS TO BE
 C ATTEMPTED. THE THREE SUBPROGRAMS ARE ...

80 C 1) SUBROUTINE F(L,K,U,UX,UXX,VAL,NPDE,IC,XSKT,XSKH)
 C DIMENSION U(NPDE),UX(NPDE),UXX(NPDE)
 C THIS ROUTINE DEFINES THE DESIRED PARTIAL DIFFERENTIAL
 C EQUATIONS TO BE SOLVED. THE PACKAGE PROVIDES VALUES OF THE
 C INPUT SCALARS L AND X AND INPUT ARRAYS (LENGTH NPDE) U, UX,
 C AND UXX, AND THE USER MUST CONSTRUCT THIS ROUTINE TO COMPUTE
 C THE OUTPUT ARRAY VAL (LENGTH NPDE) WHICH CONTAINS THE
 C CORRESPONDING VALUES OF THE "RIGHT HAND SIDES OF THE DESIRED
 C PARTIAL DIFFERENTIAL EQUATIONS. I.e.

85 C FVAL(K) = THE VALUE OF THE RIGHT HAND SIDE OF THE K-TH PDE IN
 C THE PDE SYSTEM A(U), FOR K = 1 TO NPDE.

90 C THE INCOMING VALUE OF THE SCALAR QUANTITY X WILL BE A
 C COLLOCATION POINT VALUE (SEE INITIAL AND COLPNT) AND THE
 C INCOMING VALUES IN THE ARRAYS U, UX AND UXX CORRESPOND TO THIS
 C POINT X AND TIME T.
 C RETURN END

95 C 2) SUBROUTINE ANDRY(L,T,X,U,UX,UXX,UZIN,INPUT)
 C DIMENSION U(NPDE),UX(NPDE),UXX(NPDE)
 C DIMENSION UHUX(NPDE),UHUXX(NPDE)
 C THIS ROUTINE IS USED TO PROVIDE THE PDE PACKAGE WITH NECESSARY
 C INFORMATION ABOUT THE BOUNDARY CONDITION FUNCTIONS H AND Z
 C ABOVE. THE PACKAGE PROVIDES VALUES OF THE INPUT VARIABLES
 C T, X, U, AND UX, AND THE USER IS TO DEFINE THE CORRESPONDING
 C OUTPUT VALUES OF THE DERIVATIVES OF THE FUNCTIONS H AND Z
 C WHERE ...

100 C DHUX(K)=PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF H
 C WITH RESPECT TO U(K).
 C DHUXX(K)=PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF H
 C WITH RESPECT TO UX(K).
 C DZDT(K)=DERIVATIVE OF THE K-TH COMPONENT OF Z
 C WITH RESPECT TO T.

105 C

76/76 OPT=1 MOUND=-*/ TRACE FIN 4.8049H 04/15/80 11:10:37 PAGE 4

C THE FOLLOWING CORRESPONDING VALUES OF THE OUTPUT ARRAYS
 C UDUU, UDUX, AND UDUXX FOR K,J = 1 TO NPUT...
 C UDUU(K,J) = PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF THE
 C PDE DEFINING FUNCTION F WITH RESPECT TO THE
 C VARIABLE U(J).
 C UDUX(K,J) = PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF THE
 C PDE DEFINING FUNCTION F WITH RESPECT TO THE
 C VARIABLE UX(J).
 C UDUXX(K,J) = PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF THE
 C PDE DEFINING FUNCTION F WITH RESPECT TO THE
 C VARIABLE UX(J).
 C NOTE... THE INCOMING VALUE OF X WILL BE A COLLOCATION POINT
 C VALUE.
 C RETURN
 C END

C METHODS USED
 C
 C THE PACKAGE PDECOL IS BASED ON THE METHOD OF LINES AND USES A
 C FINITE ELEMENT COLLOCATION PROCEDURE (WITH PIECEWISE POLYNOMIALS
 C AS THE TRIAL SPACE) FOR THE DISCRETIZATION OF THE SPATIAL VARIABLE
 C X. THE COLLOCATION PROCEDURE REDUCES THE PDE SYSTEM TO A SEMI-
 C DISCRETE SYSTEM WHICH THEN DEPENDS ONLY ON THE TIME VARIABLE T.
 C TIME INTEGRATION IS THEN ACCOMPLISHED BY USE OF SLIGHTLY
 C MODIFIED STANDARD TRUNCATIONS (SEE REFS. 1,2).

C PIECEWISE POLYNOMIALS
 C
 C THE USER IS REQUIRED TO SELECT THE PIECEWISE POLYNOMIAL SPACE
 C WHICH IS TO BE USED TO COMPUTE HIS APPROXIMATE SOLUTION. FIRST, THE
 C UNDER, KNUD, OF THE POLYNOMIALS TO BE USED MUST BE SPECIFIED
 C (KNUD = POLYNOMIAL DEGREE + 1). NEXT, THE NUMBER OF PIECES
 C (INTERVALS), NINT, INTO WHICH THE SPATIAL DOMAIN (LEFT,XRIGHT) IS
 C TO BE DIVIDED, IS CHOSEN. THE MIN + 1 DISTINCT BREAKPOINTS OF
 C THE DOMAIN MUST BE DEFINED AND SET INTO THE ARRAY XKPT IN
 C STRICTLY INCREASING ORDER, I.E.
 C XLEFT=XKPT(1) * LT, XKPT(2) * LT, ..., LT, XKPT(NINT)=XRIGHT.
 C THE APPROXIMATE SOLUTION AT ANY TIME T WILL BE A POLYNOMIAL OF
 C ORDER KNUD OVER EACH SUBINTERVAL (XKPT(I),XKPT(I+1)). THE
 C NUMBER OF CONTINUITY CONDITIONS, NCC, TO BE IMPOSED ACROSS ALL OF
 C THE BREAKPOINTS IS THE LAST PIECE OF USE SUPPLIED DATA WHICH IS
 C REQUIRED TO UNIQUELY TERMINATE THE DESIRED PIECEWISE POLYNOMIAL
 C SPACE. FOR EXAMPLE, NCC = 2 WOULD REQUIRE THAT THE APPROXIMATE
 C SOLUTION (MADE UP OF THE SEPARATE POLYNOMIAL PIECES) AND ITS FIRST
 C SPATIAL DERIVATIVE BE CONTINUOUS AT THE BKT APPOINTS AND WHICH ON
 C THE ENTIRE DOMAIN (LEFT,XRIGHT). NCC = 3 WOULD REQUIRE THAT THE
 C APPROXIMATE SOLUTION AND ITS FIRST AND SECOND SPATIAL DERIVATIVES,
 C BE CONTINUOUS AT THE BKT APPOINTS. ETC. THE DIMENSION OF THIS LINEAR
 C SPACE IS KNOWN AND FINITE AND IS NPTS = KNUD(NINT-1).
 C THE WELL-KNOWN H-SPLINE BASIS (SER REF.) FOR THIS SPACE IS LISTED
 C BY PDECOL AND IT CONSISTS OF NCP's KNOWN PIECEWISE POLYNOMIAL
 C FUNCTIONS HFI(I,X). FOR I=1 TO NCP, WHICH DO NOT DEPEND ON THE
 C TIME VARIABLE T, WE WISH TO EMPHASIZE THAT THE PIECEWISE POLYNOMIAL
 C SPACE LISTED IN PDECOL WHICH IS SELF-CONTAINED WILL NOT CONTAIN
 C THE MAGNITUDE OF THE SPATIAL DISCRETIZATIONS BCKUPS IN THE COMPUTED
 C APPROXIMATE SOLUTION. THE PACKAGE HAS NO CONTROL OVER THE PDE'S

76/76 OUT=1 MOUNT=-0 / FACT FIN 4.04495 04/15/80 11-10-37 PAGE 5

INTRODUCED BY THE USER'S CHOICE OF THIS SPACE. ST INPUT PARAMETER, PDECUL PDFCOL 230
 HELM. PDFCOL 231
 COLLOCATION OVER PIECEWISE POLYNOMIALS PDFCOL 232
 THE BASIC ASSUMPTION MADE IS THAT THE APPROXIMATE SOLUTION PDFCOL 233
 SATISFIES PDFCOL 234

$$U(T,X) = \sum_{i=1}^{NPTS} C(i,T) * HF(i,X)$$
 PDFCOL 235
 WHERE THE UNKNOWN COEFFICIENTS C DEPEND ONLY ON THE TIME T AND PDFCOL 236
 THE KNOWN BASIS FUNCTIONS DEPEND ONLY ON X (WE HAVE ASSUMED THAT PDFCOL 237
 NPUE = 1 FOR CONVENIENCE). SO, AT ANY GIVEN TIME T THE APPROXIMATE PDFCOL 238
 SOLUTION IS A PIECEWISE POLYNOMIAL IN THE USER CHOSEN SPACT. THE SEMI-DISCRETE EQUATIONS (ACTUALLY ORDINARY DIFFERENTIAL EQUATIONS) WHICH DETERMINE THE COEFFICIENTS C ARE OBTAINED BY PDFCOL 239
 REQUIRING THAT THE ABOVE APPROXIMATE U(T,X) SATISFY THE BC AND PDFCOL 240
 BOUNDARY CONDITIONS EXACTLY AT A SET OF NCPUS COLLOCATION POINTS PDFCOL 241
 (SEE CULPNT). THUS, PDECUL ACTUALLY COMPUTES THE BASIS FUNCTION PDFCOL 242
 COEFFICIENTS RATHER THAN SPECIFIC APPROXIMATE SOLUTION VALUES. PDFCOL 243
 REFERENCES PDFCOL 244
 1. MADSEN, N.K. AND R.F. SINGCOEL. PDECUL - COLLOCATION SOFTWARE PDFCOL 245
 FOR PARTIAL DIFFERENTIAL EQUATIONS. ACM-TUMS. VOL. 1 NO. 1. PDFCOL 246
 2. SINGCOEL, R.F. AND N.K. MADSEN. SOFTWARE FOR NONLINEAR PARTIAL PDFCOL 247
 DIFFERENTIAL EQUATIONS. ACM-TUMS. VOL. 1. NO. 3. PDFCOL 248
 SEPTEMBER 1975. PP. 232-260. PDFCOL 249
 3. MINDMARSH, A.C. PRELIMINARY DOCUMENTATION OF GEARIH, A SOLUTION PDFCOL 250
 OF IMPLICIT SYSTEMS OF ORDINARY DIFFERENTIAL EQUATIONS WITH PDFCOL 251
 HANDED JACOBIANS. LAWRENCE LIVERMORE LAB. UCID-30130. FEBRUARY PDFCOL 252
 1976. PDFCOL 253
 4. UTHOUK, C. PACKAGE FOR CALCULATING WITH B-SPLINES. SIAM J. PDFCOL 254
 NUMER. ANAL. VOL. 14. NO. 3. JUNE 1977. PP. 41-47. PDFCOL 255
 PDFCOL 256
 USE OF PDECUL PDFCOL 257
 PDECUL IS CALLED ONCE FOR EACH DESIRED OUTPUT VALUE (TOUT) OF THE PDFCOL 258
 TIME T, AND IT IN TURN MAKES REPEAT CALLS TO THE CORE SUBROUTINE PDFCOL 259
 STIFIN, WHICH ADVANCES THE TIME BY TAKING SINGLE STEPS UNTIL PDFCOL 260
 T IS TOUT. INTERPOLATION TO THE EXACT TIME TOUT IS THEN DONE. PDFCOL 261
 SEE TOUT BELOW.
 270
 SUMMARY OF SUGGESTED INPUT VALUES
 IT IS OF COURSE IMPOSSIBLE TO SUMMARIZE ALL INPUT PARAMETERS WHICH PDFCOL 271
 ARE APPROPRIATE FOR ALL PROBLEMS. THE FOLLOWING SUGGESTIONS ARE TO PDFCOL 272
 BE USED ONLY IF YOU HAVE NO IDEA OF WHAT VALUES FOR YOUR PROBLEM. PDFCOL 273
 PDFCOL 274
 OUT = 1.0-10 PDFCOL 275
 XLEFT = CHOOSE MINIMUM X-LEFT VALUE SUCH THAT XLEFT <= PDFCOL 276
 XRIGHT = XLEFT AND XRIGHT = XRIGHT. PDFCOL 277
 TOUT = 1.0-4 PDFCOL 278

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C NINT = ENOUGH SO THAT ANY FINE STRUCTURE IN THE PROBLEM MAY BE
C RESOLVED.          PDECOL 247
C NRU = 4             PDECOL 248
C NCC = 2             PDECOL 249
C MF = 2^2            PDECOL 250
C INITX = 1 (ON FIRST CALL ONLY. INITX 0 THEREAFTER). PDECOL 251
C                                     PDECOL 252
C                                     PDECOL 253
C                                     PDECOL 254
C                                     PDECOL 255
C                                     PDECOL 256
C                                     PDECOL 257
C                                     PDECOL 258
C                                     PDECOL 259
C                                     PDECOL 260
C                                     PDECOL 261
C                                     PDECOL 262
C                                     PDECOL 263
C                                     PDECOL 264
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C                                     PDECOL 266
C                                     PDECOL 267
C                                     PDECOL 268
C                                     PDECOL 269
C                                     PDECOL 270
C                                     PDECOL 271
C                                     PDECOL 272
C                                     PDECOL 273
C                                     PDECOL 274
C                                     PDECOL 275
C                                     PDECOL 276
C                                     PDECOL 277
C                                     PDECOL 278
C                                     PDECOL 279
C                                     PDECOL 280
C                                     PDECOL 281
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C                                     PDECOL 339
C                                     PDECOL 340
C                                     PDECOL 341
C                                     PDECOL 342
C                                     PDECOL 343

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THE INPUT PARAMETERS ARE:

T0 = THE INITIAL VALUE OF T. THE INDEPENDENT VARIABLE
 (USED ONLY ON FIRST CALL).

TOUT = THE VALUE OF T AT WHICH INPUT IS DESIRED NEXT. SINCE
 THE PACKAGE CHOOSES ITS OWN TIME STEP SIZES, THE
 INTEGRATION WILL NORMALLY GO SLIGHTLY BEYOND TOUT
 AND THE PACKAGE WILL INTERPOLATE TO T = TOUT.

UT = THE INITIAL STEP SIZE IN T. IF INDEX = 1 OR THE
 MAXIMUM STEP SIZE ALLOWED (MUST BE .07. OR. IF INITX = .5.
 USED FOR INPUT ONLY WHEN INDEX = 1 OR 3. SEE BELOW).

XKPT = THE ARRAY OF PIECEWISE POLYNOMIAL HAT APPROXIMATIONS.
 THE NINT+1 VALUES MUST BE STRICTLY INCREASING, WITH
 XKPT(1) = LEFT AND XKPT(NINT+1) = RIGHT (USED ONLY
 ON FIRST CALL).

EPS = THE RELATIVE TIME ERROR (USED ONLY ON THE
 FIRST CALL, UNLESS INITX = 4). SINGLE STEP ERTHK
 ESTIMATES DIVIDED BY CMAX(1) WILL BE KEPT LESS THAN
 EPS IN ROOT-PEAN-SQUART NORM. THE VECTOR CMAX OF WEIGHTS
 IS COMPUTED IN PDECUL. INITIALLY CMAX(1) IS SET TO
 ABS(C(1)) WITH A DEFAULT VALUE OF 1 IF ABS(C(1)) < 1.
 THEREAFTER, CMAX(1) IS THE LARGEST VALUE
 OF ABS(C(I)) SEEN SO FAR, OR THE INITIAL CMAX(1) IF
 THAT IS LARGER. TO AVOID EITHER OF THESE, CHANGE THE
 APPROPRIATE STATEMENTS IN THE SUBROUTINES ENDING AT
 STATEMENTS 50 AND 130 "LOW. THE USER SHOULD EXERCISE
 SOME DISCRETION IN CHOOSING EPS. IN GENERAL, THE
 OVERALL RUNNING TIME FOR A PROBLEM WILL BE GREATER IF
 EPS IS CHOSEN SMALLER. THERE IS USUALLY LITTLE PAYOFF TO
 CHOOSE EPS MUCH SMALLER THAN THE TOLERANCES WHICH ARE SET IN
 INTRODUCED BY THE USER'S CHOICE OF THE POLYNOMIAL SPACE.
 SEE RELATED COMMENTS CONCERNING CMAX BELOW STATEMENT 40.

NINT = THE NUMBER OF SUBINTERVALS INTO WHICH THE SPATIAL DOMAIN
 (LEFT,XRIGHT) IS TO BE DIVIDED (MUST BE .GF. 1)
 (USED ONLY ON FIRST CALL).

KNUD = THE ORDER OF THE PIECEWISE POLYNOMIAL SPACE TO BE USED.
 ITS VALUE MUST BE GREAT THAN 2 AND LESS THAN 21. FOR
 FIRST ATTEMPTS WE RECOMMEND KNUD = 4. IF THE SOLUTION
 IS SMOOTH AND MUCH ACCURACY IS DESIRED, HIGHER VALUES
 MAY PROVE TO BE MORE EFFICIENT. WE HAVE SET KNUD USE 9
 VALUES OF KNUD IN EXCESS OF NINT 9, THOUGH THEY ARE
 AVAILABLE FOR USE IN PDECUL (USED ONLY ON FIRST CALL).

NCC = THE NUMBER OF CONTINUITY CONDITIONS TO BE IMPOSED ON THE
 APPROXIMATE SOLUTION AT THE INTERPOLANTS IN XKPT(1).
 NCC MUST BE GREATER THAN 1 AND LESS THAN KNUD. WE
 RECOMMEND THE USE OF NCC = 2 (WITH ERTHK = 0.5).

INDEX = SINCE THEORY PREDICTS THAT ORGANICALLY MORE
 ACCURATE RESULTS CAN OBTAIN IF THE COMPUTE IS INITIATED USING THIS CHOICE
 (USED ONLY ON FIRST CALL).

NPOINT = THE NUMBER OF PARTIAL DIFFERENTIAL EQUATIONS IN THE SYSTEM

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C MF = TO HE SOLVED (USED ONLY ON FIRST CALL).
 C C INDEX = 4). ALLOWED VALUES ARE 11, 12, 21, 22.
 C FOR FIRST ATTEMPT WE COMMENT THE USE OF MF = 22.
 C MF HAS TWO DECIMAL DIGITS. METH AND MITEH
 C MF = LUENIH + MITEH.
 C METH IS THE BASIC METHOD INDICATOR.
 C METH = 1 MEANS THE ADAMS METHODS (GENERALIZATIONS OF
 C CRANK-NICOLSON).
 C METH = 2 MEANS THE NICKMAKU DIFFERENTIATION
 C FORMULAS (LUFF) IN STIFF METHODS OF GAW.
 C MITEH IS THE ITERATION METHOD INDICATOR
 C AND DETERMINES HOW THE JACOBIAN MATRIX IS
 TO BE COMPUTED.
 C MITEH = 1 MEANS CHUMI METHOD WITH ANALYTIC JACOBIAN.
 C FOR THIS USER SUPPLIES SUBROUTINE INTIVE.
 C SEE DESCRIPTION ABOVE.
 C MITEH = C MEANS CHUMI METHOD WITH JACOBIAN CALCULATION
 C INTERNALLY BY FINITE DIFFERENCES. SET
 C SUBROUTINE PSETIN AND UIFFF.
 C MITEH = INTEGER USED ON INPUT TO INDICATE TYPE OF CALL.
 C WITH THE FOLLOWING VALUES AND MEANINGS.
 C 1 THIS IS THE FIRST CALL FOR THIS PROBLEM.
 C 0 THIS IS NOT THE FIRST CALL FOR THIS PROBLEM.
 C AND INTEGRATION IS TO CONTINUE.
 C SAME AS 0 EXCEPT THAT TOUT IS TO BE HIT
 C EXACTLY (NO INTERPOLATION IS DONE!). SET NOTE
 C BELOW. ASSUME TOUT = 0. THE CURRENT T.
 C IF TOUT IS .LT. THE CURRENT TIME, THEN TOUT IS
 C SET TO THE CURRENT TIME AND CONTROL IS
 C RETURNED TO THE USER. A CALL TO VALUES WILL
 C PRODUCE ANSWERS FOR THE NEW VALUE OF TOUT.
 C SAME AS 0 EXCEPT CONTROL RETURNS TO CALLIN,
 C WHICH AFTER ONE STEP. TOUT IS IGNORED AND
 C IT MUST BE SET .GT. 0 TO A MAXIMUM ALLOWED
 C DT VALUE. SEE ABOVE.
 C 4 THIS IS NOT THE FIRST CALL FOR THE PROBLEM.
 C AND THE USER HAD REQUESTED ANY OTHER MF.
 C SINCE THE NORMAL OUTPUT VALUE OF IOUT IS 0,
 C IT NEED NOT BE HIT FOR NORMAL CONTINUATION.

C NOTE: THE PACKAGE MUST HAVE TAKEN AT LEAST ONE SUCCESSFUL TIME
 C STEP BEFORE A CALL WITH INDEX = 2 OR 4 IS ALLOWED.
 C AFTER THE INITIAL CALL, IF A NORMAL RETURN OCCURRED AND A NORMAL
 C CONTINUATION IS DESIRED, SIMPLY RESET TOUT AND CALL AGAIN.
 C ALL OTHER PARAMETERS WILL BE READY FOR THE NEXT CALL.
 C A CHANGE OF PARAMETERS WITH INDEX = 0 CAN HIT. MAYBE AFTER
 C EITHER A SUCCESSFUL OR AN UNSUCCESSFUL RETURN PROVIDED AT LEAST
 C ONE SUCCESSFUL TIME STEP HAS BEEN MADE.

C MITEH = FLOATING POINT WORKING ARRAY FOR PDECOL. WE RECOMMEND YOU
 C THAT IT BE INITIALIZED TO ZERO BEFORE THE FIRST CALL
 C TO PDECOL. ITS TOTAL LENGTH MUST BE AT LEAST

C C MITEH = LWORK + ICPTSO (300...1000) *
 C C MITEH = ICPTSO (300...1000) *
 C C MITEH = ICPTSO (300...1000) *

344 PDFCOL
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 400 PDFCOL

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400
C WHEN M1 AND MAXUW ARE DEFINED BELOW (SEE STORAGE
C ALLOCATION).
C INTEGER WORKING ARRAY FOR PDECUL. THE FIRST TWO
C LOCATIONS MUST BE USED AS FOLLOWS...
C IWORK(1) = LENGTH OF WORK ARRAY WORK
C IWORK(2) = LENGTH OF WORKS ARRAY IWORK
C THE TOTAL LENGTH OF IWORK MUST BE AT LEAST
C NCPTS*(INPUT + 1).
C
C OUTPUT
C
410
C THE SOLUTION VALUES ARE NOT RETURNED DIRECTLY TO THE USER BY PDECUL.
C THE METHODS USED IN PDECUL COMPUTE THIS FUNCTION COEFFICIENTS. SO
C THE USER (AFTER A RETURN FROM PDECUL) MUST CALL THE PACKAGE ROUTINE
C VALUES TO OBTAIN HIS APPROXIMATE SOLUTION VALUES AT ANY DESIRED SPACE
C POINTS X AT TIME T = TOUT. SEE THE COMMENTS IN SUBROUTINE VALUES.
C FOR DETAILS ON HOW TO PROPERLY MAKE THE CALL.
C
C THE COMMON BLOCK /GEAR0/ CAN BE ACCESSED EXTERNALLY BY THE USER
C IF DESIRED. IT CONTAINS THE STEP SIZE LAST USED (SUCCESSFULLY).
C THE OTHER LAST USED (SUCCESSFULLY). THE NUMBER OF STEPS TAKEN
C SO FAR. THE NUMBER OF RESIDUAL EVALUATIONS (THE CALLS) SO FAR.
C AND THE NUMBER OF MATRIX EVALUATIONS (PSETIM CALLS) SO FAR.
C DIFFUN CALLS ARE COUNTED IN WITH RESIDUAL EVALUATIONS.
C
C THE OUTPUT PARAMETERS ARE...
C
417
C DT = THE STEP SIZE USED LAST. WHETHER SUCCESSFULLY OR NOT.
C TOUT = THE OUTPUT VALUE OF T. IF INTEGRATION WAS SUCCESSFUL.
C AND THE INPUT VALUE OF INDEX WAS NOT 3, TOUT IS
C UNCHANGED FROM ITS INPUT VALUE. OTHERWISE, TOUT
C IS THE CURRENT VALUE OF T TO WHICH THE INTEGRATION
C HAS BEEN COMPLETED.
C
420
C INUTX = INTEGER USED ON OUTPUT T, INDICATE RESULTS.
C
C
421
C 0          WITH THE FOLLOWING VALUES AND MEANINGS...
C -1          INTEGRATION WAS COMPLETED TO TOUT OR HYDNU.
C             THE INTEGRATION WAS HALTED AFTER FAILING TO PASS THE
C             ERROR TEST EVEN AFTER REDUCING IT BY A FACTOR OF
C             1.E10 FROM ITS INITIAL VALUE.
C             AFTER SOME INITIAL SUCCESS. THE INTEGRATION WAS
C             HALTED EITHER BY REPEATED ERROR TEST FAILURES OR BY
C             A TEST ON TPS. TOO MUCH ACCURACY HAS BEEN REQUESTED.
C             THE INTEGRATION WAS HALTED AFTER FAILING TO ACHIEVE
C             CORRECTOR CONVERGENCE EVEN AFTER REDUCING IT BY A
C             FACTOR OF 1.E10 FROM ITS INITIAL VALUE.
C             SINGULAR MATRIX ENCOUNTERED. PROBABLY DUE TO STOREAGE
C             OVERWRITES.
C
424
C -5          INDEX WAS 4 ON INPUT. BUT THE DESIRED CHANGES OF
C             PARAMETERS WERE NOT IMPLEMENTED BECAUSE TOUT
C             WAS NOT HYDNU. IF "POLATUM 10" = TOUT WAS
C             PERFORMED AS ONE A NUMMUL. RETURN. TO TRY AGAIN,
C             SIMPLY CALL AGAIN WITH INDEX = 4 AND A NEW TOUT.
C             ILLEGAL INDEX VALUE.
C
427
C -7          ILLEGAL EPS' VALUE.
C             AN ATTEMPT TO INITIATE IN THE WORKS DIRECTLY. THE
C             SIGN OF UT IS WRONG, RETURN TO TRY AGAIN.
C             DT = 0.0.
C             ILLEGAL INITIAL VALUE.
C
430
C -10         ILLEGAL KUMU VALUE.
C             -11

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76/76  OPT=1 MOUNU=--> INACT   FIN 4.04.64H    04/15/H0  11.10.37  DATA

C -12  ILLEGAL NCL VALUE.          PDECOL 448
C -13  ILLEGAL INPUT VALUE.        PDECOL 449
C -14  ILLEGAL MF VALUE.          PDECOL 450
C -15  ILLEGAL BREAKPOINTS WHICH STRICTLY INCREASING.  PDECOL 451
C -16  INSUFFICIENT STORAGE FOR WORK ON IWORK.          PDECOL 452
C
C SUMMARY OF ALL PACKAGE ROUTINES
C PDECOL - STORAGE ALLOCATION, ERROR CHECKING, INITIALIZATION, REINITIALI-
C CALLS TO STIFF TO ADVANCE THE TIME.
C
C INTEHP - INTERPOLATES COMPUTED BASIS FUNCTION COEFFICIENTS TO THE
C DESIRED OUTPUT TIMES, TOUT, FOR USE BY VALUES.
C
C INITIAL - INITIALIZATION, GENERATION AND STORAGE OF PIECEWISE POLY-
C NOMIAL SPACE BASIS FUNCTION VALUES AND DERIVATIVES. DETER-
C MINES THE BASIS FUNCTION COEFFICIENTS OF THE PIECEWISE
C POLYNOMIALS WHICH INTERPOLATE THE USER'S INITIAL CONDITIONS.
C
C CULPNT - GENERATION OF REQUIRED COLLOCATION POINTS.
C
C HSPLNU - H-SPLINE PACKAGE ROUTINES WHICH ALLOW FOR EVALUATION OF
C HSPLVN ANY H-SPLINE BASIS FUNCTION, OR DERIVATIVE VALUE.
C
C INITWV
C
C VALUES - GENERATION AT ANY POINT(S) (IF VALUES OF THE COMPUTED)
C APPROXIMATE SOLUTION AND ITS DERIVATIVES WHICH ARE
C PIECEWISE POLYNOMIALS. THIS SUBROUTINE IS CALLED ONLY BY
C THE USER.
C
C STIFFH - CORE INTEGRATION, TAKES SINGLE TIME STEPS TO ADVANCE THE
C TIME, ASSEMBLES AND SOLVES THE PROPER NONLINEAR EQUATIONS
C WHICH ARE RELATED TO USE OF ADAMS OR GEAR TYPE INTEGRATION
C FORMULAS, CHOOSES PROPER STEP SIZE AND INTEGRATION FORMULA
C ORDER TO MAINTAIN A DESIRED ACCURACY. DESIGNED FOR ODE
C PROBLEMS OF THE FORM A * (Y,T) = G(T,Y).
C
C CUST1 - GENERATES INTEGRATION FORMULA AND THERM CONTROL COEFFICIENTS.
C
C HTS - COMPUTES RESIDUAL VECTORS USED IN SOLVING THE NONLINEAR
C EQUATIONS BY A MODIFIED NEWTON METHOD.
C
C DIFUN - COMPUTES A**1 * G(T,Y) WHERE A AND G ARE AS ABOVE (STIFFH).
C
C AUUA - ADDS THE A MATRIX TO A GIVEN MATRIX IN HAND FORM.
C
C FVAL - EVALUATES THE COMPUTED PIECEWISE POLYNOMIAL SOLUTIONS AND
C DERIVATIVES AT COLLOCATION POINTS.
C
C GFUN - EVALUATES THE FUNCTION G(T,Y) BY CALLING EVAL AND THE USER
C SUBROUTINES F AND HMDP.
C
C PST1 - GENERATES PIECEWISE JACOBIAN MATRIX, RECALCULATED BY THE
C NEWTON METHOD.
C
C

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76/76 OPT=1 ROUND=--*/ TRACT FTN 4.0.44H 04/15/80 11.10.37 PAGE 10

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C DIFFF - PERFORMS SAME MUL AS THE OTHER ROUTINE DERIVE. COMPUTE'S
C DERIVATIVE APPROXIMATIONS BY USE OF FINITE DIFFERENCES.
C DUCH - PERFORM AN LU DECOMPOSITION AND FORWARD AND BACKWARD
C SOLN SUBSTITUTION FOR SOLVING Banded SYSTEMS OF LINEAR EQUATIONS.
C
C
C STORAGE ALLOCATION
C
C SINCE PDECUL IS A DYNAMICALLY DIMENSIONED PROGRAM, MOST OF ITS
C WORKING STORAGE IS PROVIDED BY THE USER IN THE ARRAYS WORK AND IWORK.
C THE FOLLOWING GIVES A LIST OF THE ARRAYS WHICH MAKE UP THE CURRENTS
C WORK AND IWORK. THEIR LENGTHS, AND THEIR USES. WHEN MORE THAN ONE
C NAME IS GIVEN, IT INDICATES THAT DIFFERENT NAMES ARE USED FOR THE
C SAME ARRAY IN DIFFERENT PARTS OF THE PROGRAM. THE DIFFERENT NAMES
C OCCUR BECAUSE PDECUL IS AN AMALGAMATION OF SEVERAL OTHER CODES.
C WRITTEN BY DIFFERENT PEOPLE AND WE HAVE TRIED TO LEAVE THE SEPARATE
C PARTS AS UNCHANGED FROM THEIR ORIGINAL VERSIONS AS POSSIBLE.
C
C
C NAMES LENGTH USE
C ----- -----
C CC_HL 4*NPDF HOURS CONDITION INFORMATION
C WORK(IW1) 3*KOKOU*NCPTS BASIS FUNCTION VALUES AT COLLOCATION POINT
C WORK(IW2) NCPTS * KOKU BREAKPOINT SEQUENCE FOR GENERATION OF BASIS
C XC_XI NCPTS * KOKU FUNCTION, VALUES.
C XC_WOK(W3) NCPTS COLLOCATION POINTS.
C CMAX NPDF*NCPTS VALUES USED IN ESTIMATING TIME
C YMAX
C WORK(IW4) NPDF*NCPTS INTEGRATION ERRORS.
C FWORK NPDF*NCPTS TIME INTEGRATION ERRORS.
C WORK(IW5) NPDF*NCPTS WORKING STORAGE FOR TIME INTEGRATION
C SAVE1 NPDF*NCPTS METHOD.
C WORK(IW6) NPDF*NCPTS WORKING STORAGE FOR TIME INTEGRATION
C SAVE2 NPDF*NCPTS METHOD.
C WORK(IW7) NPDF*NCPTS WORKING STORAGE FOR TIME INTEGRATION
C SAVE3 NPDF*NCPTS METHOD.
C WORK(IW8) NPDF*NCPTS WORKING STORAGE FOR TIME INTEGRATION
C UVAL 3*NPDF WORKING VALUES OF U(X) AT
C WORK(IW9) NPDF*NCPTS CURRENT ASIS FULLITION COEFFICIENT VALUE.
  
```


76/76 OPT=1 ROUND=0-0/ 1-ACE FIN 4.H4YH 04/15/80 11.10.37 PAt 12
 C AS INDICATED BELOW.
 C PACKAGE ROUTINES CALLED..
 C IISDP ROUTINES CALLED..
 C CALLED BY..
 C FORTHAN FUNCTIONS USED..
 C

```

630      DIMENSION WORK(11),IWORK(11),THKPT(11)
          COMMON /GEAR0/ UTUSED,NOUSED,NSHP,NFE,NJE,
          COMMON /GEAK1/ TUTC,DTMC,DTNU,SC,UHOU,NU,MFC,KFLAG,JSTAK
          COMMON /GEAR9/ EPSJ,RUML,MU,MW,N11,NUML,JU,WNUW
          COMMON /OPTION/ NODAUS,MAXDTK
          COMMON /SIZES/ NI,NJ,KUR,NC,NPU,NPTS,NQD,IQUD
          COMMON /START/ IW1,IW2,IW3,IW4,IW5,IW6,IW7,IW8,IW9,IW10,
          COMMON /LUNIT/ LUUT
          IF (INDEX .EQ. 0) GO TO 60
          IF (INDEX .EQ. 2) GO TO 70
          IF (INDEX .EQ. 4) GO TO 80
          IF (INDEX .EQ. 3) GO TO 90
  
```

C SEVERAL CHECKS ARE MADE HERE TO DETERMINE IF THE INPUT PARAMETERS
 C HAVE LEGAL VALUES. ERROR CHECKS ARE MADE ON INITX, EPS, (TU-TOUT)*NU,
 C DI, NINT, KORD, NCC, NPD. MF WHETHER THE BREAKPOINT SEQUENCE IS
 C STRICTLY INCREASING AND WHETHER THERE IS SUFFICIENT STORAGE
 C PROVIDED FOR WORK AND IWORK. PROBLEMS DEPENDENT PARAMETERS ARE
 C CALCULATED AND PLACED IN COMMON.

```

635      IERID = -6
          IF (INDEX .NE. 1) GO TO 320
          IERID = IFRID - 1
          IF (EPS .LE. 0.) GO TO 320
          IERID = IFRID - 1
          IF ((TU-TOUT)*NU .GT. 0.) GO TO 420
          IERID = IFRID - 1
          IF (DT .EQ. 0.0) GO TO 320
          IERID = IFRID - 1
          NIN = NINT
          IF (NIN .LT. 1) GO TO 320
          IERID = IFRID - 1
          KUR = KORD
          IF (KUR .LT. 3 .OR. KUR .GT. 40) GO TO 320
          IERID = IFRID - 1
          NC = NCC
          IF (NCC .LT. 2 .OR. NCC .GT. NIN) GO TO 320
          IERID = IFRID - 1
          NPD = NPDF
          NPDE2 = NPD*NPD
          IF (NPDE .LT. 1) GO TO 320
          IERID = IFRID - 1
          IF (MF .NE. 22 .AND. MF .NE. 21 .AND. MF .NE. 12 .AND. MF .NE. 11) GO TO 320
          IERID = IFRID - 1
          DO 10 K=1,NIN
  
```

640 CONTINUE
 10 NPTS = KUR * (1.0 - 1) * (KUR - NC)
 NCUN = NPDF * NPTS

645 PDECUL 629
 PDECUL 630
 PDECUL 631
 PDECUL 632
 PDECUL 633
 PDECUL 634
 PDECUL 635
 PDECUL 636
 PDECUL 637
 PDECUL 638
 PDECUL 639
 PDECUL 640
 PDECUL 641
 PDECUL 642
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 PDECUL 674
 PDECUL 675
 PDECUL 676
 PDECUL 677
 PDECUL 678
 PDECUL 679
 PDECUL 680
 PDECUL 681
 PDECUL 682
 PDECUL 683
 PDECUL 684

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645      ML=KOH-2
       MU = ML
       MW = ML + ML + 1
       NU=NUMP*(3*ML+1)
       NW1=NCPTS*(3*ML+1)
       IWSAVE = IWORK(1)
       IWSAVE = IWORK(2)
       IWI=4*NPUF+1
       IW2 = IW1 + 3*KORU*NCPTS
       IW3 = IW2 + NCPTS * KORU
       IW4 = IW3 + NCPTS
       IW5 = IW4 + NEUN
       IW6 = IW5 + NEQN
       IW7 = IW6 + NEQI
       IW8 = IW7 + NEQN
       IW9 = IW8 + NEQN
       IW10 = IW9 + 3*NPUE
       IW11 = IW10 + NEUN*(MAXDER+1)
       IW12=IW1+NPUE
       IW13=IW12+NPUDE
       IW14=IW13+NPUE
       IW15=IW14+NPUE
       IW16=IW15+NPUE
       IW17 = IW16 + NPUE
       IW18 = NCPTS + 1
       ITRID = IEIDU - 1
       IWSOK = IW17 + NEUN*(3*ML+1) - 1
       IWSOK = IW18 + NEUN - 1
       IF(IWSAVE .LT. IWSOK .OR. IWSAVE .LT. IWSOK ) GO TO 335
C-----  

C PERFORM INITIALIZATION TASKS. IF KOID .EQ. 3 THEN CALCULATE THE HAD-J-  

C WIDTH OF THE ASSOCIATED MATRIX PROBLEM BY DETERMINING THE TYPE OF  

C BOUNDARY CONDITIONS. THEN CHECK FOR SUFFICIENT STORAGE AGAIN.
C-----  

C-----  

       CALL INITIAL(KOH,WORK(IWI),WORK(IW1),XKHP1,WORK(IW2),WORK(IW3)).  

*      WORK(IWI),WORK(IW1),WORK(IW2),WORK(IW3))  

646      IF (ILOAD .NE. 0) GO TO 200
       IWSOK = IW17 + NEUN*(3*ML+1) - 1
       IF (IWSAVE .LT. IWSOK ) GO TO 335
C-----  

C IF INITIAL VALUES OF CMAX OTHER THAN THOSE SET BELOW ARE DESIRED,  

C THEY SHOULD BE SET HERE. ALL CMAX() MUST BE POSITIVE.  

C HAVING PROPER VALUES OF CMAX FOR THE PROBLEM BEING SOLVED IS AS  

C IMPORTANT AS CHOOSING EPS (SEE AHNU). STRICT ERRORS ARE  

C MEASURED RELATIVE TO CMAX. IF VALUES FOR UMIN OR UMAX, THE  

C ROUNDS ON ABS(IY) OTHER THAN THOSE BELOW ARE DESIRED, THEY  

C SHOULD BE SET BELOW.
C-----  

C SEE C MODIFIES THE ERROR CONTROL.
C AN ABSOLUTE ERROR CRITERION IS USED FOR UNINITIATES LESS THAN .001.
       DU 50 1 = 1*NEUN
       WORK(IWI+1)=AMAX1(ABS(WORK(IWI+1)),.001)
647      DU  WORK(IWI+1) = WORK(IWI+1)
       IY = NY-IY
       I = I0
       OTC = I1

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14          PA5T          11.10.37          04/15/HU          FIN 4.0.H-44H

SUBROUTINE PDECUL      7677A      (PNT=1) WOUMU=--*--/  TRACT      FIN 4.0.H-44H

743          UTMN = ABS(DT)
    DTUSED = 0.
    EPSC = EPS
    MFC = MF
    JSTART = 0
    EP SJ = SORT(UKUNUNU)
    NH1 = NEON - 1
    NH2L = NEON*NL
    NHCLUT >= 0
    KFLAG = 0
    TU1P = T0
    IF ( T0 .EQ. TU1P ) GO TO 360
    DU UTMX = ABS(TOUT-TU1P)*10.
    GO TO 140

750          C   70 UTMX = ABS(TOUT-TU1P)*10.
    IF ((T-TU1P)*UTC .GT. 0.) GO TO 340
    GO TO 150

755          C   80 IF ((T-TU1P)*DTC .GT. 0.) GO TO 300
    JSTART = -1
    EPSC = EPS
    MFC = MF
    GO TO 100

760          C   90 STIFIR(NEON*TU1P*WORK(1W1), WORK(1W4)*WUWK(1W5)*WOK(1W6),
    *           WORK(1W7)*WUWK(1W8)*WUWK(1W17)*WUWK(1W18)*WOK(1W9))
    *           WORK(1W19)

765          C   90 UTMX = DT
    100 IF ((T+DTC) .EQ. T) WHITE(LOU1*110)
    110 FORMAT(36H WARNING.. 1 + DT = 1 ON NEXT STEP.)
    C----- TAKE A TIME STEP BY CALLING THE INITRATCH.
    C----- CALL STIFIR(NEON*TU1P*WORK(1W1), WORK(1W4)*WUWK(1W5)*WOK(1W6),
    *           WORK(1W7)*WUWK(1W8)*WUWK(1W17)*WUWK(1W18)*WOK(1W9)
    *           WORK(1W19))

770          C   KGU = 1 - KFLAG
    GO TO 120, 160, 220, 260, 2HU1, KGU
    C   KFLAG = 0, -1, -2, -3, -4
    C   120 CONTINUE
    C----- NORMAL RETURN FROM INTEGRATOR.
    C----- THE WEIGHTS CMX(1) ARE UPDATED.  IF DIFFERENT VALUES ARE INPUTTED,
    C----- THEY SHOULD BE SET HERE.  A TEST IS MADE FOR THIS BEING TOO SMALL
    C----- FOR THE MACHINE PRECISION.

775          C----- ANY OTHER TESTS OR CALCULATIONS THAT ARE REQUIRED AFTER EVERY
    C----- STEP SHOULD BE INSERTED HERE.

780          C   IF (INDX = 3) SAVEL IS SET TO THE CURRENT C VALUES, UN RETURN.
    C   IF (INDX = 2, DT IS COMPRESSED TO HIS TOUT LARGEST MOUNDIF F
    C   EHUKU). AND THEN THE CURRENT C VALUES ARE PUT IN SAV1 ON HIS TOUT.
    C   FOR ANY OTHER VALUE OF INDEX, CINTML RETURNS TO THE INITRATCH IF
    C   UNLESS TOUT HAS BEEN REACHED. THEN INTERPOLATED VALUES OF C ARE
    C   COMPUTED AND STORED IN SAVEL OR RETURN.
    C   IF INTERPOLATION IS NOT DESIRED. HIS CALL TO INITRACH IS READ OF
    C   REMOUNT AND CONTROL TRANSFERRED TO STATEMENT 140.

785          C----- CUMOUNT AND CONTROL TRANSFERRED TO STATEMENT 140 HIS READ OF
    C----- 360.

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SUMROUTINT PROBLIN 76/76 0HPT=1 HOUND=-*-*/ 1rACT FIN 4.0.H+4UH 04/15/H0 11.10.37 PAGE 15

R00      U = 0.                                     H00
        DU 130 I = 1. NEGUN                      PDECUL
        LI = 1 - 1                                PDECUL
        AYI = AHS(MWORK(L1W0+1))                  PDECUL
        NMWK(L1W4+1) = AMAX(L1WEC,AYI)           PDECUL
        130 U = I * (AYI/MWK(L1W4+1))**2          PDECUL
        U = D*(IROUND/EPS)**2                     PDECUL
        IF (U > .6T) FLOAT(1EUNI) GO TU 400      PDECUL
        IF (INDEX .EQ. J) GU TO 340              PDECUL
        IF (INDEX .EQ. 2) GU TO 150              PDECUL
        140 IF (I1-TOUT)*DTC .LT. 0.) GU TU 100    PDECUL
        CALL INTERP(TOUT,MWK(L1W0)*NU(.0)*WK(L1W5)) PDECUL
        GO TO 360                                H12

C      150 IF (((IT*DTC)-TOUT)*DTIC .LE. 0.) GO TU 100
        IF (ABS(IT-TOUT) .LT. 100.*NU(.0)*DTMK) GO TU 340
        IF ((IT-TOUT)*DTC .LT. 0.) GO TU 440
        DTIC = (TOUT - IT)*(1. - 4.*NU(.0))
        JSTART = -1
        GO TO 100                                H13

C      ON AN ERROR RETURN FROM INTEGRATOR. AN IMMEDIATE RETURN OCCURS IF
C      KFLAG = -2 OR -4 AND RECOVERY AT IT'S ARE MADE OTHERWISE.
C      TO RECOVER, DT AND DTWN ARE REDUCED BY A FACTOR OF *1 UP TO 10
C      TIMES BEFORE GIVING UP.
C-----H21

R15      160 WRITE (ILOUT,170) IT
        170 FORMAT 1/35H KFLAG = -1 FROM INTEGRATOR AT IT = *E16.8/
        * 40H ERROR TEST FAILED WITH N,S(UIT) = DIMNS)
        180 IF (NMHCUT .EQ. 10) GO TO 200
        NMHCUT = NMHCUT + 1
        DTWN = 1.0*DTMN
        DTIC = 1.0*DTC
        WHITE (ILOUT,190) UTC
        190 FORMAT (25H DT HAS BEEN REDUCED TO *E16.8)
        * 26H AND STEP WILL BE RETRIVED
        JSTART = -1
        GO TO 100                                H22

C      200 WRITE (ILOUT,*210)
        210 FORMAT 1//44H PROBLEM APPEARS UNRESOLVABLE WITH GIVEN INPUT//)
        GO TO 340                                H23

C      220 WRITE (ILOUT,*230) T,DTIC
        230 FORMAT 1//35H KFLAG = -2 FROM INTEGRATOR AT IT = *E16.8/H DT =
        * E16.8/22H THE REQUESTED ERROR IS SMALLER THAN CAN HF HANDLE IT //
        GO TO 340                                H24

C      240 WRITE (ILOUT,*250) T
        250 FORMAT 1//37H INTEGRATION HALTED BY DIVIDE AT IT = *E16.8/H
        * 56H EPS TOO SMALL TO HF ATTENDED FOR INT. MACHINE PUT(CLISION)
        KFLAG = -2
        GO TO 340                                H25

R40      H40
R45      H45
R46      H46
R47      H47
R48      H48
R49      H49
R50      H50
R51      H51
R52      H52
R53      H53
R54      H54
R55      H55
R56      H56
R57      H57

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SUBROUTINE P1COL      76/76   OPT=1 ROUND=--*/ TRACE          FIN 4.0.H44H    04/15/80 11.10.37 PAGE  16
*
* 45H CORRECTION CONVENIENCE WOULD NOT BE ACHIEVED/
* GO TO 180
C 290 WRITE (IOUT,290)
290 FORMAT(//28H SINGULAR MATRIX ENCOUNTERED.
* 35H PRIMARILY DUE TO STORAGE OVERWITES//)
NFLAG = -4
GO TO 340
C 300 WRITE (IOUT,310) T,TOUT,DTIC
310 FORMAT(//4SH INITX = -1 ON INPUT WITH (1-TOUT)*D1 .GT. 0./
* 4H T = E16.89H TOUT = E16.89H DTIC = E16.89/
* 44H INTERPOLATION WAS DONE AS ON NORMAL RETURN.//
* 41H DESIRED PARAMETER CHANGES WERE NOT MADE.|)
CALL INTERP(TOUT,WORK(110),NEW(110))
INDEX = -5
RETURN
C 320 WRITE (IOUT,330) ITWU
330 FORMAT(//24H ILLEGAL INPUT...INDEX = 13//)
INDEX = IERID
RETURN
C 335 WRITE (IOUT,336) LNSIZH,LNSAVE,JJ,TOH,LLSAWE
336 FORMAT(//21H INSUFFICIENT STOCHT/24H WORK MUST BE OF LENGTH
* 110*5K12H YOU PROVIDED 110/24H WORK MUST BE OF LENGTH,110,*X.
* 12H YOU PROVIDED,110//)
INDX = IERID
RETURN
C 340 TUUT = 1
00 350 I = 1.NE.ON
11 = 1
350 WORK(1W6+11) = WORK(1W10+11)
360 INDEX = KFLAU
TUUTP = TINIT
UF = OTUSED
IF (KFLAU .NE. 0) DI = DTIC
RETURN
END

```

```

1      SUBROUTINE USOL(X,USOL,SCTCH,N1,N2,N3,N4,N5,N6,N7,N8,N9)
2      VALUES
3      VALUES
4      VALUES
5      VALUES
6      VALUES
7      VALUES
8      VALUES
9      VALUES
10     VALUES
11     VALUES
12     VALUES
13     VALUES
14     VALUES
15     VALUES
16     VALUES
17     VALUES
18     VALUES
19     VALUES
20     VALUES
21     VALUES
22     VALUES
23     VALUES
24     VALUES
25     VALUES
26     VALUES
27     VALUES
28     VALUES
29     VALUES
30     VALUES
31     VALUES
32     VALUES
33     VALUES
34     VALUES
35     VALUES
36     VALUES
37     VALUES
38     VALUES
39     VALUES
40     VALUES
41     VALUES
42     VALUES
43     VALUES
44     VALUES
45     VALUES
46     VALUES
47     VALUES
48     VALUES
49     VALUES
50     VALUES
51     VALUES
52     VALUES
53     VALUES
54     VALUES
55     VALUES
56     VALUES
57     VALUES
58     VALUES
59     VALUES
60     VALUES
61     VALUES
62     VALUES
63     VALUES
64     VALUES
65     VALUES
66     VALUES
67     VALUES
68     VALUES
69     VALUES
70     VALUES
71     VALUES
72     VALUES
73     VALUES
74     VALUES
75     VALUES
76     VALUES
77     VALUES
78     VALUES
79     VALUES
80     VALUES
81     VALUES
82     VALUES
83     VALUES
84     VALUES
85     VALUES
86     VALUES
87     VALUES
88     VALUES
89     VALUES
90     VALUES
91     VALUES
92     VALUES
93     VALUES
94     VALUES
95     VALUES
96     VALUES
97     VALUES
98     VALUES
99     VALUES
100    VALUES
101   -----
102   C THE CALLING PARAMETERS ARE:**
103   C   X = AN ARBITRARY VECTOR OF SPATIAL POINTS OF LENGTH NPTS AT
104   C   WHICH THE SOLUTION AND THE FIRST ORDER DERIVATIVE VALUES
105   C   ARE TO BE CALCULATED. IF X .LT. XLEFT ( X .GT. XRIGHT )
106   C   THEN THE PIECEWISE POLYNOMIAL OVER THE LEFTMOST ( RIGHT-
107   C   MOST ) INTERVAL IS EVALUATED TO CALCULATE THE SOLUTION
108   C   VALUES AT THIS UNUSUAL VALUE OF X. SEE PDECOL.
109
110   C USOL = AN ARRAY WHICH CONTAINS THE SOLUTION AND THE FIRST
111   C ORDER DERIVATIVES OF THE SOLUTION AT ALL THE POINTS IN
112   C THE INPUT VECTOR X. IN PARTICULAR, USOL(J,I) CONTAINS
113   C THE VALUE OF THE (K-1)-ST DERIVATIVE OF THE J-TH PDE
114   C COMPONENT AT THE I-TH POINT OF THE X VECTOR FOR
115   C J = 1 TO NPUT. I = 1 TO NPTS. AND K = 1 TO NDERV+1.
116
117   C SCTCH = A USER SUPPLIED WORKING STORAGE ARRAY OF LENGTH AT LEAST
118   C NUDV*(NDEV+1). SEE HELV1 AND PUECUL FOR DEFINITIONS OF
119   C THESE PARAMETERS.
120
121   C N1,N2 = THE FIRST DIMENSION OF THE OUTPUT ARRAY IN THE CALLING
122   C PROGRAM. NUDV MUST BE .LE. NPUT.
123
124   C N1,N2 = THE SECOND DIMENSION OF THE OUTPUT ARRAY USOL IN THE
125   C CALLING PROGRAM. NUDV MUST BE .LE. NPTS.
126
127   C NPTS = THE NUMBER OF POINTS IN THE X VECTOR.
128
129   C NDERV = THE NUMBER OF DERIVATIVE VALUES OF THE SOLUTION THAT ARE
130   C TO BE CALCULATED. NDERV SHOULD BE LESS THAN NUDV SINCE
131   C THE KUDV-TH DERIVATIVE OF A POLYNOMIAL OF DEGREE KUDV-1
132   C IS EQUAL TO ZERO. SET PUDCOL.
133
134   C NUDV = THE USERS WORKING STORAGE ARRAY WHICH IS USED IN THIS CALL.
135   C TO PROVIDE THE CURRENT HISTORY FUNCTION COEFFICIENTS AND THE
136   C PIECEWISE POLYNOMIAL HISTOPOINT SEQUENCE.
137
138   C PACKAGE ROUTINES CALLED... HSPLV1,L1,ENV
139   C USER ROUTINES CALLED... NONE
140   C CALLED BY... USERS MAIN PROGRAM
141   C FUNCTIONS USED... NONE
142
143   C -----
144   C DIMENSION USOL(NUDV+1,NDEV+1),X(NPTS),SCTCH(1),WORK(1)
145   C COMMON /START/ NUDV,KUDV,NCPLT,NCPTR,NUTV,INVAL
146   C DATA ILFT/0/,MLAY/0/
147   C NUTV1 = NDERV + 1
148   C DO 20 IPTS=1,NPTS
149   C   CALL INTFHV(WORK(IPTS),NUTV1,X(IPTS),ILFT,MLAY)
150

```

SUBROUTINE VALUES 7676 OPT=1 ROUND=-*/ 1-ACT F7N 4.0.H+4YH 04/15/80 11.10.37 PAG 14
 CALL HSPLVU(WORK(1W2)*K(0,0,0,A(1,1S)),(Lft,I,SCrch,NodeRv))
 IK = ILFFT - K(0,0)
 DO 10 M=1,NodeRv
 II = (M-1)*K(0,0)
 DO 10 K=1,NPUE
 USOL(K,IPTS,M) = 0.
 IS=IW6-1+(K-1)*IPTS+IK
 DO 10 I=1,K(0,0)
 USOL(K,IPTS,M)=USOL(K,IPTS,M)+W(I,K((IS+I)*SCrch(I+II)))
 10 CONTINUE
 20 CONTINUE
 HF TUPN
 END

60 67 70

VALUES 59
 VALUES 60
 VALUES 61
 VALUES 62
 VALUES 63
 VALUES 64
 VALUES 65
 VALUES 66
 VALUES 67
 VALUES 68
 VALUES 69
 VALUES 70
 VALUES 71

```

1          SUBROUTINE INIT(K,A,RHS,X,T,A,P,W,IPIV,L,F1)
2
3          C INITIAL IS CALLED ONLY ONCE BY PNTCOL TO PERFORM INITIALIZATION TASKS.
4          C THESE TASKS INCLUDE - 1) DEFINING THE PIECEWISE POLYNOMIAL SPACE
5          C BREAKPOINT SEQUENCE. 2) CALLING THE SUBROUTINE COLPNT TO DEFINE THE
6          C REQUIRED COLLOCATION POINTS. 3) DEFING THE PIECEWISE POLYNOMIAL SPACE.
7          C BASIS FUNCTION VALUES (PLUS FIRST AND SECOND DERIVATIVE VALUES) AT
8          C THE COLLOCATION POINTS, AND 4) DEFINING THE INITIAL BASIS FUNCTION
9          C COEFFICIENTS WHICH DETERMINE THE PIECEWISE POLYNOMIAL WHICH
10         C INTERPOLATES THE USER SUPPLIED (INIT) INITIAL CONDITION FUNCTIONS(S)
11         C AT THE COLLOCATION POINTS.
12
13         C K = ORDER OF PIECEWISE POLYNOMIAL SPACE.
14         C A = BASIS FUNCTION VALUES GENERATED BY INITIAL.
15         C RHS = TEMPORARY STORAGE USED TO RETURN INITIAL CONDITION CUEST+IC1T.
16
17         C X = USER DEFINED PIECEWISE POLYNOMIAL BREAKPOINT VALUES.
18         C XT = PIECEWISE POLYNOMIAL BREAKPOINT SEQUENCE GENERATED BY INITIAL.
19         C XC = COLLOCATION POINTS GENERATED BY INITIAL.
20         C PW = STORAGE FOR BAND MATRIX USED TO GENERATE INITIAL
21         C COEFFICIENT VALUES.
22         C IPIV = PIVOT INFORMATION FOR LINEAR EQUATION SOLVER DECH-SULH.
23         C ILTFT = POINTERS TO BREAKPOINT SEQUENCE GENERATED BY INITIAL.
24
25         C PACKAGE ROUTINES CALLED... HSPLVD,COLPNT,DECH,INTERV,SOLH
26         C USER ROUTINES CALLED... UINIT
27         C CALLD HY...
28         C FURTHER FUNCTIONS USED... PDECOL
29         C MAX0,MIN0
30
31         DIMENSION A(K,3,1),RHS(1),X(1),AT(1,1),XC(1,1),PW(1,1),IPIV(1,1),ILTFT(1)
32         COMMON/SITES/NINT,KORD,NC,NPI,NCPTS,NRNU,IER
33         COMMON/GEAR9/EP SJ,RU,ML,MU,IUUM(3),NOM,NUM
34         COMMON/HURNK/HURNK
35         COMMON/THRM/THRM
36         COMMON/TAHCF/RL,CPMX,HO(20),R2D(10),R2D(10)
37         COMMON/TAHFM/FM
38         COMMON/START/NSTART,NRURN,KRD
39         COMMON/END/IEH
40         CFLAG = -2
41         IEH = 0
42
43         C SET UP THE PIECEWISE POLYNOMIAL SPACE, BREAKPOINT SEQUENCE.
44
45         KRPD = KORD - NCL
46         DO 10 I=1,KORD
47         AT(NCPTS+1) = X(NINT+1)
48         10     X(I) = X(1)
49         DO 20 I=2,NINT
50             I1 = (I-2)*KRPD + KRPD
51             20     X(I+J) = X(I)
52
53         C SET UP COLLOCATION POINTS ARRAYS X.
54
55         CALL COLPNT(X,FC,K)
56
57         C COMPUTE THE LEFT ARRAY. STORE THE BASIS FUNCTION VALUES IN THE
58

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SUBROUTINE INITIAL      76/76   OPT=1  ROUND=--*/ TRACE          FTN 4.8+49H    04/15/80  11.10.37  WAIT    20

C ARRAY A. THE ARRAY A IS DIMENSIONED ARR(0,0,3*NCPTS) AND ARR(K,J,I)      INITIAL      59
C CONTAINS THE VALUE OF THE (J-1)-ST DERIVATIVE (J = 1..2..3) OF THE K-th      INITIAL      60
C NONZERO BASIS FUNCTION (K = 1.. ..NROW) AT THE I-th CULLOCATION      INITIAL      61
C POINT (I = 1.. ..NCPTS). SET UP LHS FOR INTERPOLATING THE INITIAL      INITIAL      62
C CONDITIONS AT THE CULLOCATION POINTS. SET THE INTERPOLATION MATRIX      INITIAL      63
C INTO THE HANDED MATRIX PH.      INITIAL      64
C
C
C DO 35 I=1,NCPTS
CALL INTERV(X(I),NCPTS,XC(I),1L+T(I),MFLAG)      INITIAL      65
CALL BSPLVO(X(I),XH(I),XC(I),1L+T(I),A(1,1,1)+3)      INITIAL      66
CONTINUE      INITIAL      67
C
35 IF INSTANT = 1*      INITIAL      68
CC THE STARTING VALUES FOR THE C'S ARE JUST THE INCOMING Y VALUES.      INITIAL      69
CC RATHER THAN SOLVING THE SYSTEM OF EQUATIONS.      INITIAL      70
CC THIS PREVENTS THE SYSTEM FROM OVERWORKING TRYING TO GET AROUND      INITIAL      71
CC SHARP CORNERS OR GENERATING NEGATIVE VALUES.      INITIAL      72
CC NM=NCPTS-1      INITIAL      73
DO 50 KPU=1,NUDE      INITIAL      74
DO 40 I=1,NCPTS      INITIAL      75
IP=I+(KPU-1)*NCPTS      INITIAL      76
CALL UNIT(XC(I)+KHS(IP),NUDE,KPDE)      INITIAL      77
IF INSTANT.EQ.1J60 TU 50      INITIAL      78
C
50 IF NMURN = 1, WE NEED THE CORRECT VALUE FOR N2U AT THE LEFT      INITIAL      79
C BOUNDARY TO GET THE PROPER MASS FLUX FRACTION.      INITIAL      80
IF (NMURN.EQ.1)GU TU <30      INITIAL      81
IF (KPDE.GT.1)GO TO 230      INITIAL      82
HEAD(1,220)(H2D(LI),L=1,NUDE)      INITIAL      83
C
230 FORMAT(PRE14.6)      INITIAL      84
WRITE(3,225)(R2I(L),L=1,NUDE)      INITIAL      85
FORMAT(1/2X,5HR2) = .1F6E14.6/)      INITIAL      86
CONTINUE      INITIAL      87
DO 30 I=1,NOW1      INITIAL      88
PU(I)=0.0      INITIAL      89
DO 3K 1=1,NM      INITIAL      90
11=I-1      INITIAL      91
ICOL=ILEFT(I,1)-1-1      INITIAL      92
JL=MAKO(I,1+2-NCPTS)      INITIAL      93
JU=MINO(KORD,KORD)+1-c)      INITIAL      94
DO 3K J=JL,JU      INITIAL      95
J1=I+NCPTS*(ICOL,J-1)      INITIAL      96
PU(I,J1)=A(J,J1)      INITIAL      97
IF (NMURN.EF.0)GU TO JR      INITIAL      98
IF (I.EQ.1.AND.KPDE.LT.NUDE)PU(I,J1)=      INITIAL      99
* A((J,1,1)-R2D(KPDE)*A(J,2,1)/FM      INITIAL      100
CONTINUE      INITIAL      101
C
100 JR
C
101 IF (NMURN.EF.0)RE TURN      INITIAL      102
102 ICOL=ILEFT(NCPTS)-NCPTS-1      INITIAL      103
DO 39 J=2,KORD      INITIAL      104
J1=NM+NCPTS*(ICOL,J-1)      INITIAL      105
PU(I,J1)=A(J,2,1)      INITIAL      106
103 KPDE=NCPTS*KPDE      INITIAL      107
104 NM=NM+NCPTS      INITIAL      108
105 NM=IP=0.0      INITIAL      109
CALL DECH(NCPTS,NCP1S,M1,M2,PW=1,IV=1ER)      INITIAL      110
IF (IWH.NE.0)RE TURN      INITIAL      111
106 I=I+1      INITIAL      112
107 NCPTS=1      INITIAL      113
108 CALL SOLV(NCPTS,NCP1S,M1,M2,PW=1,S(1,N))=1*1ER      INITIAL      114
109 CONTINUE      INITIAL      115

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SUBROUTINE INITIAL 767K OPT=1 ROUND=***/1,ACCT
RETURN TBL
115

FIN 4.049H 04/15/80 11:10:37 PAIR 21

INITIAL
116
INITIAL
117

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1      SUBROUTINE COLPNT 76/76   OPT=1 MOUNU=---/ PLACE   FIN 4.0449H  04/15/80  11.10.37
2
3      SUBROUTINE COLPNT(XC,XI)
4
5      C COLPNT IS CALLED ONLY ONCE BY INITIAL TO DEFINE THE REQUIRED COLLUCA-
6      C TION POINTS WHICH ARE TO BE USED WITH THE USER SELECTED PIECEWIS-
7      C E POLYNOMIAL SPACE. THE COLLOCATION POINTS ARE CHOSEN SUCH THAT THEY
8      C ARE EITHER THE POINTS AT WHICH THE PIECEWISE POLYNOMIAL SPACE HAS ITS
9      C FUNCTIONS ATTAIN THEIR UNIQUE MAXIMUM VALUES OR THE GAUSS-LEGENDRE
10     C QUADRATURE POINTS WITHIN EACH PIECEWISE POLYNOMIAL SPACE SUBINTERVAL.
11     C DEPENDING UPON THE SPACE BEING USED AND THE DESIRE OF THE USER.
12
13     XC = USER DEFINED PIECEWISE POLYNOMIAL BREAKPOINTS.
14     XC = COLLOCATION POINTS DEFINED BY COLPNT.
15     XI = PIECEWISE POLYNOMIAL BREAKPOINT SEQUENCE.
16
17     PACKAGE ROUTINES CALLED.. HSPLVU,I,IEHV
18     USER ROUTINES CALLED.. NONE
19     CALLED BY.. NONE
20     COMMON /FUNCTIONS USE/.. INITIAL
21     COMMON /SIZES/ NINT,KURD,NCC,NMNP,NCPTS,MEMN,ICQUAD
22     COMMON /OPTION/ NOGAUS,MAXDER
23     DATA LEFT/0/
24
25     IF THE VARIABLE NOGAUS IN THE COMMON HLUCK /OPTION/ IS SET .EU. 1.
26     THE USE OF THE GAUSS-LEGENDRE POINTS IS PROHIBITED FOR ALL CASES.
27     NOGAUS IS CURRENTLY SET .EU. 0 BY A DATA STATEMENT IN THE RLUCK DATA.
28     THE USER MAY CHANGE THIS AS DESIRED.
29
30     IF I NCC .NE. 2 .OR. NOGAUS .EQ. 1 GO TO 200
31
32     COMPUTE THE COLLOCATION POINTS TO BE AT THE GAUSS-LEGENDRE POINTS IN
33     EACH PIECEWISE POLYNOMIAL SPACE SUBINTERVAL. THE ARRAY RHO IS SET TO
34     CONTAIN THE GAUSS-LEGENDRE POINTS FOR THE STANDARD INTERVAL (-1,1).
35
36     IPTS = KURD - 2
37     GO TO (10,20,30,40,50,60,70,80,90,100,110,120,130,140,150,160,170,
38     * 180),IPTS
39     10 RHO(1) = 0.
40     GO TO 190
41     20 RHO(12) = .5773502691M9626E-00
42     RHO(11) = - RHO(12)
43     RHO(10) = 0.
44     GO TO 190
45     30 RHO(13) = .339981035H4556E-00
46     RHO(12) = - RHO(13)
47     RHO(14) = .861136311444053E-00
48     RHO(11) = - RHO(14)
49     GO TO 190
50     40 RHO(13) = .33998101056A3E-00
51     RHO(12) = - RHO(13)
52     RHO(15) = .9061794593H664E-00
53     RHO(14) = - RHO(15)
54
55     50 RHO(13) = .33998101056A3E-00
56     RHO(12) = - RHO(13)
57     RHO(15) = .9061794593H664E-00
58     RHO(14) = - RHO(15)
59
60     60 RHO(13) = .33998101056A3E-00
61     RHO(12) = - RHO(13)
62     RHO(16) = .9061794593H664E-00
63     RHO(15) = - RHO(16)
64
65     70 RHO(13) = .33998101056A3E-00
66     RHO(12) = - RHO(13)
67     RHO(17) = .9061794593H664E-00
68     RHO(16) = - RHO(17)
69
70     80 RHO(13) = .33998101056A3E-00
71     RHO(12) = - RHO(13)
72     RHO(18) = .9061794593H664E-00
73     RHO(17) = - RHO(18)
74
75     90 RHO(13) = .33998101056A3E-00
76     RHO(12) = - RHO(13)
77     RHO(19) = .9061794593H664E-00
78     RHO(18) = - RHO(19)
79
80     100 RHO(13) = .33998101056A3E-00
81     RHO(12) = - RHO(13)
82     RHO(20) = .9061794593H664E-00
83     RHO(19) = - RHO(20)
84
85     110 RHO(13) = .33998101056A3E-00
86     RHO(12) = - RHO(13)
87     RHO(21) = .9061794593H664E-00
88     RHO(20) = - RHO(21)
89
90     120 RHO(13) = .33998101056A3E-00
91     RHO(12) = - RHO(13)
92     RHO(22) = .9061794593H664E-00
93     RHO(21) = - RHO(22)
94
95     130 RHO(13) = .33998101056A3E-00
96     RHO(12) = - RHO(13)
97     RHO(23) = .9061794593H664E-00
98     RHO(22) = - RHO(23)
99
100    140 RHO(13) = .33998101056A3E-00
101    RHO(12) = - RHO(13)
102    RHO(24) = .9061794593H664E-00
103    RHO(23) = - RHO(24)
104
105    150 RHO(13) = .33998101056A3E-00
106    RHO(12) = - RHO(13)
107    RHO(25) = .9061794593H664E-00
108    RHO(24) = - RHO(25)
109
110    160 RHO(13) = .33998101056A3E-00
111    RHO(12) = - RHO(13)
112    RHO(26) = .9061794593H664E-00
113    RHO(25) = - RHO(26)
114
115    170 RHO(13) = .33998101056A3E-00
116    RHO(12) = - RHO(13)
117    RHO(27) = .9061794593H664E-00
118    RHO(26) = - RHO(27)
119
120    180 RHO(13) = .33998101056A3E-00
121    RHO(12) = - RHO(13)
122    RHO(28) = .9061794593H664E-00
123    RHO(27) = - RHO(28)
124
125    190 RHO(13) = .33998101056A3E-00
126    RHO(12) = - RHO(13)
127    RHO(29) = .9061794593H664E-00
128    RHO(28) = - RHO(29)
129
130    200 RHO(13) = .33998101056A3E-00
131    RHO(12) = - RHO(13)
132    RHO(30) = .9061794593H664E-00
133    RHO(29) = - RHO(30)
134
135    210 RHO(13) = .33998101056A3E-00
136    RHO(12) = - RHO(13)
137    RHO(31) = .9061794593H664E-00
138    RHO(30) = - RHO(31)
139
140    220 RHO(13) = .33998101056A3E-00
141    RHO(12) = - RHO(13)
142    RHO(32) = .9061794593H664E-00
143    RHO(31) = - RHO(32)
144
145    230 RHO(13) = .33998101056A3E-00
146    RHO(12) = - RHO(13)
147    RHO(33) = .9061794593H664E-00
148    RHO(32) = - RHO(33)
149
150    240 RHO(13) = .33998101056A3E-00
151    RHO(12) = - RHO(13)
152    RHO(34) = .9061794593H664E-00
153    RHO(33) = - RHO(34)
154
155    250 RHO(13) = .33998101056A3E-00
156    RHO(12) = - RHO(13)
157    RHO(35) = .9061794593H664E-00
158    RHO(34) = - RHO(35)
159
160    260 RHO(13) = .33998101056A3E-00
161    RHO(12) = - RHO(13)
162    RHO(36) = .9061794593H664E-00
163    RHO(35) = - RHO(36)
164
165    270 RHO(13) = .33998101056A3E-00
166    RHO(12) = - RHO(13)
167    RHO(37) = .9061794593H664E-00
168    RHO(36) = - RHO(37)
169
170    280 RHO(13) = .33998101056A3E-00
171    RHO(12) = - RHO(13)
172    RHO(38) = .9061794593H664E-00
173    RHO(37) = - RHO(38)
174
175    290 RHO(13) = .33998101056A3E-00
176    RHO(12) = - RHO(13)
177    RHO(39) = .9061794593H664E-00
178    RHO(38) = - RHO(39)
179
180    300 RHO(13) = .33998101056A3E-00
181    RHO(12) = - RHO(13)
182    RHO(40) = .9061794593H664E-00
183    RHO(39) = - RHO(40)
184
185    310 RHO(13) = .33998101056A3E-00
186    RHO(12) = - RHO(13)
187    RHO(41) = .9061794593H664E-00
188    RHO(40) = - RHO(41)
189
190    320 RHO(13) = .33998101056A3E-00
191    RHO(12) = - RHO(13)
192    RHO(42) = .9061794593H664E-00
193    RHO(41) = - RHO(42)
194
195    330 RHO(13) = .33998101056A3E-00
196    RHO(12) = - RHO(13)
197    RHO(43) = .9061794593H664E-00
198    RHO(42) = - RHO(43)
199
200    340 RHO(13) = .33998101056A3E-00
201    RHO(12) = - RHO(13)
202    RHO(44) = .9061794593H664E-00
203    RHO(43) = - RHO(44)
204
205    350 RHO(13) = .33998101056A3E-00
206    RHO(12) = - RHO(13)
207    RHO(45) = .9061794593H664E-00
208    RHO(44) = - RHO(45)
209
210    360 RHO(13) = .33998101056A3E-00
211    RHO(12) = - RHO(13)
212    RHO(46) = .9061794593H664E-00
213    RHO(45) = - RHO(46)
214
215    370 RHO(13) = .33998101056A3E-00
216    RHO(12) = - RHO(13)
217    RHO(47) = .9061794593H664E-00
218    RHO(46) = - RHO(47)
219
220    380 RHO(13) = .33998101056A3E-00
221    RHO(12) = - RHO(13)
222    RHO(48) = .9061794593H664E-00
223    RHO(47) = - RHO(48)
224
225    390 RHO(13) = .33998101056A3E-00
226    RHO(12) = - RHO(13)
227    RHO(49) = .9061794593H664E-00
228    RHO(48) = - RHO(49)
229
230    400 RHO(13) = .33998101056A3E-00
231    RHO(12) = - RHO(13)
232    RHO(50) = .9061794593H664E-00
233    RHO(49) = - RHO(50)
234
235    410 RHO(13) = .33998101056A3E-00
236    RHO(12) = - RHO(13)
237    RHO(51) = .9061794593H664E-00
238    RHO(50) = - RHO(51)
239
240    420 RHO(13) = .33998101056A3E-00
241    RHO(12) = - RHO(13)
242    RHO(52) = .9061794593H664E-00
243    RHO(51) = - RHO(52)
244
245    430 RHO(13) = .33998101056A3E-00
246    RHO(12) = - RHO(13)
247    RHO(53) = .9061794593H664E-00
248    RHO(52) = - RHO(53)
249
250    440 RHO(13) = .33998101056A3E-00
251    RHO(12) = - RHO(13)
252    RHO(54) = .9061794593H664E-00
253    RHO(53) = - RHO(54)
254
255    450 RHO(13) = .33998101056A3E-00
256    RHO(12) = - RHO(13)
257    RHO(55) = .9061794593H664E-00
258    RHO(54) = - RHO(55)
259
260    460 RHO(13) = .33998101056A3E-00
261    RHO(12) = - RHO(13)
262    RHO(56) = .9061794593H664E-00
263    RHO(55) = - RHO(56)
264
265    470 RHO(13) = .33998101056A3E-00
266    RHO(12) = - RHO(13)
267    RHO(57) = .9061794593H664E-00
268    RHO(56) = - RHO(57)
269
270    480 RHO(13) = .33998101056A3E-00
271    RHO(12) = - RHO(13)
272    RHO(58) = .9061794593H664E-00
273    RHO(57) = - RHO(58)
274
275    490 RHO(13) = .33998101056A3E-00
276    RHO(12) = - RHO(13)
277    RHO(59) = .9061794593H664E-00
278    RHO(58) = - RHO(59)
279
280    500 RHO(13) = .33998101056A3E-00
281    RHO(12) = - RHO(13)
282    RHO(60) = .9061794593H664E-00
283    RHO(59) = - RHO(60)
284
285    510 RHO(13) = .33998101056A3E-00
286    RHO(12) = - RHO(13)
287    RHO(61) = .9061794593H664E-00
288    RHO(60) = - RHO(61)
289
290    520 RHO(13) = .33998101056A3E-00
291    RHO(12) = - RHO(13)
292    RHO(62) = .9061794593H664E-00
293    RHO(61) = - RHO(62)
294
295    530 RHO(13) = .33998101056A3E-00
296    RHO(12) = - RHO(13)
297    RHO(63) = .9061794593H664E-00
298    RHO(62) = - RHO(63)
299
299    540 RHO(13) = .33998101056A3E-00
300    RHO(12) = - RHO(13)
301    RHO(64) = .9061794593H664E-00
302    RHO(63) = - RHO(64)
303
304    550 RHO(13) = .33998101056A3E-00
305    RHO(12) = - RHO(13)
306    RHO(65) = .9061794593H664E-00
307    RHO(64) = - RHO(65)
308
308    560 RHO(13) = .33998101056A3E-00
309    RHO(12) = - RHO(13)
310    RHO(66) = .9061794593H664E-00
311    RHO(65) = - RHO(66)
312
312    570 RHO(13) = .33998101056A3E-00
313    RHO(12) = - RHO(13)
314    RHO(67) = .9061794593H664E-00
315    RHO(66) = - RHO(67)
316
316    580 RHO(13) = .33998101056A3E-00
317    RHO(12) = - RHO(13)
318    RHO(68) = .9061794593H664E-00
319    RHO(67) = - RHO(68)
320
320    590 RHO(13) = .33998101056A3E-00
321    RHO(12) = - RHO(13)
322    RHO(69) = .9061794593H664E-00
323    RHO(68) = - RHO(69)
324
324    600 RHO(13) = .33998101056A3E-00
325    RHO(12) = - RHO(13)
326    RHO(70) = .9061794593H664E-00
327    RHO(69) = - RHO(70)
328
328    610 RHO(13) = .33998101056A3E-00
329    RHO(12) = - RHO(13)
330    RHO(71) = .9061794593H664E-00
331    RHO(70) = - RHO(71)
332
332    620 RHO(13) = .33998101056A3E-00
333    RHO(12) = - RHO(13)
334    RHO(72) = .9061794593H664E-00
335    RHO(71) = - RHO(72)
336
336    630 RHO(13) = .33998101056A3E-00
337    RHO(12) = - RHO(13)
338    RHO(73) = .9061794593H664E-00
339    RHO(72) = - RHO(73)
340
340    640 RHO(13) = .33998101056A3E-00
341    RHO(12) = - RHO(13)
342    RHO(74) = .9061794593H664E-00
343    RHO(73) = - RHO(74)
344
344    650 RHO(13) = .33998101056A3E-00
345    RHO(12) = - RHO(13)
346    RHO(75) = .9061794593H664E-00
347    RHO(74) = - RHO(75)
348
348    660 RHO(13) = .33998101056A3E-00
349    RHO(12) = - RHO(13)
350    RHO(76) = .9061794593H664E-00
351    RHO(75) = - RHO(76)
352
352    670 RHO(13) = .33998101056A3E-00
353    RHO(12) = - RHO(13)
354    RHO(77) = .9061794593H664E-00
355    RHO(76) = - RHO(77)
356
356    680 RHO(13) = .33998101056A3E-00
357    RHO(12) = - RHO(13)
358    RHO(78) = .9061794593H664E-00
359    RHO(77) = - RHO(78)
360
360    690 RHO(13) = .33998101056A3E-00
361    RHO(12) = - RHO(13)
362    RHO(79) = .9061794593H664E-00
363    RHO(78) = - RHO(79)
364
364    700 RHO(13) = .33998101056A3E-00
365    RHO(12) = - RHO(13)
366    RHO(80) = .9061794593H664E-00
367    RHO(79) = - RHO(80)
368
368    710 RHO(13) = .33998101056A3E-00
369    RHO(12) = - RHO(13)
370    RHO(81) = .9061794593H664E-00
371    RHO(80) = - RHO(81)
372
372    720 RHO(13) = .33998101056A3E-00
373    RHO(12) = - RHO(13)
374    RHO(82) = .9061794593H664E-00
375    RHO(81) = - RHO(82)
376
376    730 RHO(13) = .33998101056A3E-00
377    RHO(12) = - RHO(13)
378    RHO(83) = .9061794593H664E-00
379    RHO(82) = - RHO(83)
380
380    740 RHO(13) = .33998101056A3E-00
381    RHO(12) = - RHO(13)
382    RHO(84) = .9061794593H664E-00
383    RHO(83) = - RHO(84)
384
384    750 RHO(13) = .33998101056A3E-00
385    RHO(12) = - RHO(13)
386    RHO(85) = .9061794593H664E-00
387    RHO(84) = - RHO(85)
388
388    760 RHO(13) = .33998101056A3E-00
389    RHO(12) = - RHO(13)
390    RHO(86) = .9061794593H664E-00
391    RHO(85) = - RHO(86)
392
392    770 RHO(13) = .33998101056A3E-00
393    RHO(12) = - RHO(13)
394    RHO(87) = .9061794593H664E-00
395    RHO(86) = - RHO(87)
396
396    780 RHO(13) = .33998101056A3E-00
397    RHO(12) = - RHO(13)
398    RHO(88) = .9061794593H664E-00
399    RHO(87) = - RHO(88)
400
400    790 RHO(13) = .33998101056A3E-00
401    RHO(12) = - RHO(13)
402    RHO(89) = .9061794593H664E-00
403    RHO(88) = - RHO(89)
404
404    800 RHO(13) = .33998101056A3E-00
405    RHO(12) = - RHO(13)
406    RHO(90) = .9061794593H664E-00
407    RHO(89) = - RHO(90)
408
408    810 RHO(13) = .33998101056A3E-00
409    RHO(12) = - RHO(13)
410    RHO(91) = .9061794593H664E-00
411    RHO(90) = - RHO(91)
412
412    820 RHO(13) = .33998101056A3E-00
413    RHO(12) = - RHO(13)
414    RHO(92) = .9061794593H664E-00
415    RHO(91) = - RHO(92)
416
416    830 RHO(13) = .33998101056A3E-00
417    RHO(12) = - RHO(13)
418    RHO(93) = .9061794593H664E-00
419    RHO(92) = - RHO(93)
420
420    840 RHO(13) = .33998101056A3E-00
421    RHO(12) = - RHO(13)
422    RHO(94) = .9061794593H664E-00
423    RHO(93) = - RHO(94)
424
424    850 RHO(13) = .33998101056A3E-00
425    RHO(12) = - RHO(13)
426    RHO(95) = .9061794593H664E-00
427    RHO(94) = - RHO(95)
428
428    860 RHO(13) = .33998101056A3E-00
429    RHO(12) = - RHO(13)
430    RHO(96) = .9061794593H664E-00
431    RHO(95) = - RHO(96)
432
432    870 RHO(13) = .33998101056A3E-00
433    RHO(12) = - RHO(13)
434    RHO(97) = .9061794593H664E-00
435    RHO(96) = - RHO(97)
436
436    880 RHO(13) = .33998101056A3E-00
437    RHO(12) = - RHO(13)
438    RHO(98) = .9061794593H664E-00
439    RHO(97) = - RHO(98)
440
440    890 RHO(13) = .33998101056A3E-00
441    RHO(12) = - RHO(13)
442    RHO(99) = .9061794593H664E-00
443    RHO(98) = - RHO(99)
444
444    900 RHO(13) = .33998101056A3E-00
445    RHO(12) = - RHO(13)
446    RHO(100) = .9061794593H664E-00
447    RHO(99) = - RHO(100)
448
448    910 RHO(13) = .33998101056A3E-00
449    RHO(12) = - RHO(13)
450    RHO(101) = .9061794593H664E-00
451    RHO(100) = - RHO(101)
452
452    920 RHO(13) = .33998101056A3E-00
453    RHO(12) = - RHO(13)
454    RHO(102) = .9061794593H664E-00
455    RHO(101) = - RHO(102)
456
456    930 RHO(13) = .33998101056A3E-00
457    RHO(12) = - RHO(13)
458    RHO(103) = .9061794593H664E-00
459    RHO(102) = - RHO(103)
460
460    940 RHO(13) = .33998101056A3E-00
461    RHO(12) = - RHO(13)
462    RHO(104) = .9061794593H664E-00
463    RHO(103) = - RHO(104)
464
464    950 RHO(13) = .33998101056A3E-00
465    RHO(12) = - RHO(13)
466    RHO(105) = .9061794593H664E-00
467    RHO(104) = - RHO(105)
468
468    960 RHO(13) = .33998101056A3E-00
469    RHO(12) = - RHO(13)
470    RHO(106) = .9061794593H664E-00
471    RHO(105) = - RHO(106)
472
472    970 RHO(13) = .33998101056A3E-00
473    RHO(12) = - RHO(13)
474    RHO(107) = .9061794593H664E-00
475    RHO(106) = - RHO(107)
476
476    980 RHO(13) = .33998101056A3E-00
477    RHO(12) = - RHO(13)
478    RHO(108) = .9061794593H664E-00
479    RHO(107) = - RHO(108)
480
480    990 RHO(13) = .33998101056A3E-00
481    RHO(12) = - RHO(13)
482    RHO(109) = .9061794593H664E-00
483    RHO(108) = - RHO(109)
484
484    1000 RHO(13) = .33998101056A3E-00
485    RHO(12) = - RHO(13)
486    RHO(110) = .9061794593H664E-00
487    RHO(109) = - RHO(110)
488
488    1010 RHO(13) = .33998101056A3E-00
489    RHO(12) = - RHO(13)
490    RHO(111) = .9061794593H664E-00
491    RHO(110) = - RHO(111)
492
492    1020 RHO(13) = .33998101056A3E-00
493    RHO(12) = - RHO(13)
494    RHO(112) = .9061794593H664E-00
495    RHO(111) = - RHO(112)
496
496    1030 RHO(13) = .33998101056A3E-00
497    RHO(12) = - RHO(13)
498    RHO(113) = .9061794593H664E-00
499    RHO(112) = - RHO(113)
500
500    1040 RHO(13) = .33998101056A3E-00
501    RHO(12) = - RHO(13)
502    RHO(114) = .9061794593H664E-00
503    RHO(113) = - RHO(114)
504
504    1050 RHO(13) = .33998101056A3E-00
505    RHO(12) = - RHO(13)
506    RHO(115) = .9061794593H664E-00
507    RHO(114) = - RHO(115)
508
508    1060 RHO(13) = .33998101056A3E-00
509    RHO(12) = - RHO(13)
510    RHO(116) = .9061794593H664E-00
511    RHO(115) = - RHO(116)
512
512    1070 RHO(13) = .33998101056A3E-00
513    RHO(12) = - RHO(13)
514    RHO(117) = .9061794593H664E-00
515    RHO(116) = - RHO(117)
516
516    1080 RHO(13) = .33998101056A3E-00
517    RHO(12) = - RHO(13)
518    RHO(118) = .9061794593H664E-00
519    RHO(117) = - RHO(118)
520
520    1090 RHO(13) = .33998101056A3E-00
521    RHO(12) = - RHO(13)
522    RHO(119) = .9061794593H664E-00
523    RHO(118) = - RHO(119)
524
524    1100 RHO(13) = .33998101056A3E-00
525    RHO(12) = - RHO(13)
526    RHO(120) = .9061794593H664E-00
527    RHO(119) = - RHO(120)
528
528    1110 RHO(13) = .33998101056A3E-00
529    RHO(12) = - RHO(13)
530    RHO(121) = .9061794593H664E-00
531    RHO(120) = - RHO(121)
532
532    1120 RHO(13) = .33998101056A3E-00
533    RHO(12) = - RHO(13)
534    RHO(122) = .9061794593H664E-00
535    RHO(121) = - RHO(12
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SUBROUTINE COLPNT 76/76 OPT=1 RHO(1)=+--*/ TFACT

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60 TO 190
60 RHO(4) = .2386191H6UH3197E-00
RHO(3) = - RHO(4)
RHO(5) = .661204JH6466245E-00.
RHO(2) = - RHO(1)
RHO(6) = .932466J14203152E-00
RHO(1) = - RHO(6)

60 TO 190
70 RHO(5) = .405845151J7397E-00
RHO(3) = - RHO(5)
RHO(6) = .7415311H5L9394E-00
RHO(2) = - RHO(6)
RHO(7) = .949107912342759E-00
RHO(1) = - RHO(7)
RHO(4) = 0.

60 TO 190
80 RHO(5) = .183434624495650E-00
RHO(4) = - RHO(5)
RHO(6) = .525532409416329E-00
RHO(3) = - RHO(6)
RHO(7) = .796666477413627E-00
RHO(2) = - RHO(7)
RHO(8) = .960289456447536E-00
RHO(1) = - RHO(8)

60 TO 190
90 RHO(5) = 0
RHO(6) = .32425342J403809E-00
RHO(7) = .61337143J700594E-00
RHO(1) = .836031107326616E-00
RHO(9) = .968160234507625E-00
DU 95 I=1,4
95 RHO(1) = -RHO(10)-1
DU 10 TO 190
100 RHO(6) = .14887433H981631F-00
RHO(7) = .633395339412924E-00
RHO(8) = .67940955H299024E-00
RHO(9) = .865063360088684E-00
RHO(10) = .97390652H517172F-00
DU 105 I=1,5
105 RHO(1) = -RHO(11)-1
DU 10 TO 190
110 RHO(6) = 0
RHO(7) = .269543155542345E-00
RHO(1) = .519096129206812E-00
RHO(9) = .730152005574049E-00
RHO(10) = .86706259974H091E-00
RHO(11) = .97822465M146057E-00
DU 115 I=1,5
115 RHO(1) = -RHO(12)-1
DU 10 TO 190
120 RHO(7) = .12523340H511469E-00
RHO(1) = .367H3149H9A1H1H-00
RHO(9) = .54731745428H617E-00
RHO(10) = .769902674L4L4L3E-00
RHO(11) = .904117250371675E-00
RHO(12) = .981560H34256719E-00
DU 115 I=1,6

```

SUBROUTINE COLPNT 76/76 OPT=1 ROUND=-0.0/ INACT

FIN 4.0 H+44H 04/15/H0 11.10.37

HAT

?4

```

115      125  MM0(1) = -MM0(13-1)
          GO TO 190
130  MM0( 7) = 0
          MM0( 8) = .230458315455145E-00
          MM0( 9) = .448492751036447E-00
          MM0(10) = .64234934440340E-00
          MM0(11) = .8015700900733310E-00
          MM0(12) = .9175949922974E-00
          MM0(13) = .98418305471854H-00
          DO 135 I=1,6
135  MM0(I) = -MM0(14-I)
          GO TO 190
140  MM0( 6) = .10805946707344E-00
          MM0( 9) = .3191126H927890E-00
          MM0(10) = .51524883581544E-00
          MM0(11) = .687298946811685E-00
          MM0(12) = .827201315069765E-00
          MM0(13) = .928434883663574E-00
          MM0(14) = .9H628H08696812E-00
          DO 145 I=1,7
145  MM0(I) = -MM0(15-I)
          GO TO 190
150  MM0( 5) = 0
          MM0( 9) = .201194093997435E-00
          MM0(10) = .39415347077563E-00
          MM0(11) = .570912172608539E-00
          MM0(12) = .72441731560170E-00
          MM0(13) = .848206583410427E-00
          MM0(14) = .93723392407076E-00
          MM0(15) = .987492518U20485E-00
          DO 155 I=1,7
155  MM0(I) = -MM0(15-I)
          GO TO 190
160  MM0( 9) = .9501250948376374E-01
          MM0(10) = .281601550779259E-00
          MM0(11) = .458016771657227E-00
          MM0(12) = .617H12444402644E-00
          MM0(13) = .75540440835503E-00
          MM0(14) = .8656312023H7832E-00
          MM0(15) = .944570123073233E-00
          MM0(16) = .989440934991659E-00
          DO 165 I=1,8
165  MM0(I) = -MM0(17-I)
          GO TO 190
170  MM0( 9) = 0
          MM0(10) = .174441H1495H44HF-00
          MM0(11) = .351231763453H76E-00
          MM0(12) = .5126915310H647E-00
          MM0(13) = .65767115216691E-00
          MM0(14) = .7H151403896801E-00
          MM0(15) = .890239153726986E-00
          MM0(16) = .9506752176H76E-00
          MM0(17) = .940571475314417E-00
          DO 175 I=1,9
175  MM0(I) = -MM0(18-I)
          GO TO 190
180  MM0(10) = .9477H01304173531E-01

```

SUBROUTINE COLPNT 7676 OPT=1 MUNDO=+0+ TRACE

04/15/80 11.10.37

PAGE 25

```

      KHO(11) = -251846227691506t.00          CULPNT 173
      KHO(12) = -41175116146284t.00          CULPNT 174
      KHO(13) = -5977043107394t.00          CULPNT 175
      KHO(14) = -6916874306035t.00          CULPNT 176
      KHO(15) = -8037044589725t.00          CULPNT 177
      KHO(16) = -892602466497556t.00          CULPNT 178
      KHO(17) = -955H2394457139Ht.00          CULPNT 179
      DO 195 I=1,9                         KHO(I) = .99156516t420931E-00
      190                                         CULPNT 180
      185   KHO(I) = -KHO(I-1)                  CULPNT 181
      C COMPUTE THE GAUSS-Legendre COLLOCATION POINTS IN EACH SUBINTERVAL.
      C
      195   DO 196 I=1,NINT
            FAC = ( X(I+1) - X(I) ) * .5
            DO 195 J = 1,IPTS
            KNOT = IPTS * (I-1) + J + 1
            XC(KNOT) = X(I) + FAC * ( X(I+1) + X(I) )
            XC(I) = X(I)
            XC(IPTS) = X(NINT+1)
            RETURN
      C COMPUTE THE COLLOCATION POINTS TO HAT AT THE POINTS WHERE THE BASIS
      C FUNCTIONS ATTAIN THEIR MAXIMA. A BISECTION METHOD IS USED TO FIND
      C THE POINTS TO MACHINE PRECISION. THIS PROCESS COULD BE SPECIFIED UP
      C BY USING A SECANT METHOD IF DESIRABLE.
      C
      200   ITUP = NCPTS - 1
      MFLAG = -2
      XC(I) = X(I)
      XC(IPTS) = X(NINT+1)
      DO 240 I=2,ITUP
      XOLD = 1.0E+20
      XL = X(I)
      XR = X(I+KORD)
      KNEW = .5 * (XL + XR)
      IF( XOLD .EQ. XNEW ) GO TO 240
      CALL INFHW(XT,KORD,XNEW,ILFT,1,MFLAG)
      CALL HSPLWD(XT,KORD,XNEW,ILFT,1,MFLAG)
      DO 220 J=1,KORD
      IF( I .EQ. J .AND. ILFT = KRD ) GO TO 230
      CONTINUE
      220   XVAL = WHO(KRD)+J
      230   IF( XVAL .EQ. 0.0 ) XR = XNEW
      IF( XVAL .GT. 0.0 ) XL = XNEW
      IF( XVAL .LT. 0.0 ) XR = XNEW
      KOLD = XNEW
      GO TO 210
      240   XC(I) = XR
      RETURN
      END

```

SUBROUTINE	COLUMN	NUMBER	SEVERITY	DETAILS
		41	1	
		44	1	
		44	1	
		50	1	
		53	1	
		55	1	
		51	1	
		63	1	
		66	1	
		64	1	
		70	1	
		76	1	
		78	1	
		80	1	
		84	1	
		85	1	
		86	1	
		87	1	
		92	1	
		93	1	
		94	1	
		95	1	
		101	1	
		102	1	
		103	1	
		104	1	
		109	1	
		110	1	
		111	1	
		112	1	
		113	1	
		119	1	
		120	1	
		121	1	
		122	1	
		123	1	
		124	1	
		129	1	
		130	1	
		131	1	
		132	1	
		133	1	
		139	1	
		140	1	
		141	1	
		142	1	
		143	1	
		144	1	
		145	1	
		149	1	
		150	1	
		151	1	
		152	1	
		153	1	
		154	1	

76/76 OUT=1 ROUND=-0.011-ACE DIAGNOSIS OF MURKOFF

FIN 4.0.66464 04/15/80 11.10.37

三

85

SUBROUTINE COLPNT

76/76 OPT=1 MOUND=-33/1-FACT

FIN 4.0044H

04/15/60 11.10.37

PACt

		DIAGNOSIS OF PROBLEM		
CARD	NR.	SEVERITY	DETAILS	
155	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
161	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
162	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
163	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
164	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
165	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
166	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
167	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
171	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
173	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
174	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
175	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
176	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
177	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
178	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.
179	1	CONSTANT TOO LONG.	HIGH UNK/R DIGITS RETAINED.	BUT SOME PRECISION LOST.

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7676 OPT=1 MOUND=-0/-1 ACT FIN 4.0.49H 04/15/80 11-10-37 PAGE 24

1      SUBROUTINE BSPLVN ( X, K, X0, L, FT, VNIKX, NUTRIV )          HSPLVU 2
2      C-- THIS SUBROUTINE IS PART OF THE H-SPLINE PACKAGE FOR THE STABLE      HSPLVU 3
3      C EVALUATION OF ANY H-SPLINE BASIS FUNCTION OR DERIVATIVE. VALUE.      HSPLVU 4
4      C SEE REFERENCE BELOW.                                              HSPLVU 5
5
6      C CALCULATES THE VALUE AND THE FIRST  $n$  DERIVATIVES OF ALL      HSPLVU 6
7      C H-SPLINES WHICH DO NOT VANISH AT X. THE ROUTINE FILLS THE TWO-      HSPLVU 7
8      C DIMENSIONAL ARRAY VNIKX(J,IODEIV). J=1DTRIV, ... ,K WITH NODERIV      HSPLVU 8
9      C VALUES OF H-SPLINES OF ORDER K+1-L-1. VNIKX(J,IODEIV), ... ,1.      HSPLVU 9
10     C REPEATED CALLS TO BSPLVN.                                         HSPLVU 10
11
12    C XT = PIECEWISE POLYNOMIAL BREAKPOINT SEQUENCE.                  HSPLVU 11
13    C K = ORDER OF THE PIECEWISE POLYNOMIAL SPACE.                      HSPLVU 12
14    C X = POINT AT WHICH THE H-SPLINE IS TO BE EVALUATED.             HSPLVU 13
15    C ILEFT = POINTER TO THE BREAKPOINT PRECEDENT.                   HSPLVU 14
16    C VNIKX = TABLE OF H-SPLINE VALUES AND DERIVATIVES.              HSPLVU 15
17    C NUTRIV = DETERMINES NUMBER OF DERIVATIVES TO BE GENERATED.       HSPLVU 16
18
19    C PREFERENCE
20
21    C UTHOR. C. package for calculating with n-splines. SIAM J.          HSPLVU 17
22    C NUMER. ANAL., VOL. 14, NO. 3, JUNE 1977, PP. 441-472.           HSPLVU 18
23
24    C PACKAGE ROUTINES CALLED.. HSPLVN
25    C USER ROUTINES CALLED..  NONE
26    C CALLED BY..          COLPNT,LITAL,VALUES
27    C FURTHER FUNCTIONS USED.. FLOAT,MAXI
28
29    C-----DIMENSION X(1),VNIKX(K,NUTRIV)
30    C-----DIMENSION A(20,20)
31    C-----K0 = K + 1 - NDERIV
32    C-----HSPLVU
33    C-----CALL BSPLVN(XT,K0,1,X,ILEFT,VNIR,(NUTRIV,NUTRIV))
34    C-----IF (NUTRIV .LE. 1) GO TO 120
35    C-----NUTRIV = NDERIV
36    C-----DO 20 I=2,NDERIV
37    C----- 1DFWRM = IODEIV-1
38    C----- 1D 10 J=1,IEHV*K
39    C----- 10  VNIKX(J-1,1DHWRM) = VNIKX(J,IODEIV)
40    C----- 1DIEHV = IUEHV
41    C----- 1CALL BSPLVN(XT,U,C,X,ILEFT,VNIR,(IEHV,1DHWRM))
42    C----- 20 CONTINUE
43    C----- 1D 40 I=1,K
44    C----- 10  J=1,K
45    C----- 10   A(I,J) = 0.
46    C----- 40   A(I,I) = 1.
47    C----- KMU = K
48    C----- 1D 110 M=2,MDFWIV
49    C----- 1D   KMU = KMU - 1
50    C----- 1D   FMU = FLOAT(KMU)
51    C----- 1   I = ILEFT
52    C----- 1   J = K
53    C----- 10   JPI = J-1
54    C----- 10   IPKD = 1 + KMU
55    C----- 10   IF (IPKD .EQ. 0) GO TO 1D
56    C----- 10   IF (IPKD .EQ. 0, 0, 1D) GO TO 1D
57    C----- 10   IF (IPKD .EQ. 0, 0, 1D) GO TO 1D

```

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SUBROUTINE HSPLV0      76/76      DMT=1  K(UN)=--> / 1*ACT
      FIN 4.0K+49H      04/15/HU  11.10.37      FMT      24

      60      100 60 L=1,J      HSPLVU      59
      60      A(L,J) = (A(L,J-1)) / DIFF * KMU
      70      J = JM      HSPLVU      60
      70      1 = 1 - 1      HSPLVU      61
      60      GO TO 50      HSPLVU      62
      80      IF (DIFF .EQ. 0.) GO TO 90      HSPLVU      63
      80      A(1,1) = A(1,1)/DIFF * KMU      HSPLVU      64
      DO 110 J=1,K      HSPLVU      65
      90      VU      HSPLVU      66
      90      DO 110 J=1,K      HSPLVU      67
      90      V = 0.      HSPLVU      68
      90      JLUM = MAX0(1,M)
      90      DO 100 J=JLUM      HSPLVU      69
      100      V = A(1,J)*VNKK(J,M) + V      HSPLVU      70
      100      VNKK(1,M) = V      HSPLVU      71
      120      RETURN      HSPLVU      72
      END      HSPLVU      73

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76/76 OPT=1 MUNIUE=*** 1-ACE FIN 4.0.H+4YN 04/15/80 11.10.37 PAIR 30

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1      SUBROUTINE BSPLVN ( X, JHIGH, IWORK, X, ILFT, VNIKX )          HSPLVN 2
2      C THIS SUBROUTINE IS PART OF THE H-SPLINE PACKAGE FOR THE STABLE
3      C EVALUATION OF ANY H-SPLINE BASIS FUNCTION OR DERIVATIVE VALUE.
4      C SEE REFERENCE BELOW.
5
6      C CALCULATES THE VALUE OF ALL POSSIBLY NONZERO H-SPLINES AT THE
7      C POINT X OF ORDER MAX(JHIGH, (J+1) (J+2)) FOR THE BREAKPOINT STRUCTURE
8      C USEN X. ASSUMING THAT X<ILFT <LE <LT. X(ILFT) IS THE KNOT
9      C INT RETURNS THE H-SPLINE VALUES IN THE ONE DIMENSIONAL ARRAY VNIKX.
10
11      C FOR DEFINITIONS OF CALLING ARGUMENTS SEE ABOVE AND HSPLVD.
12
13      C REFERENCE
14
15      C DEHOOR, C. PACKAGE FOR CALCULATION WITH H-SPLINES. SIAM J.
16      C NUMER. ANAL., VOL. 14, NO. 3, JUNE 1971, PP. 441-472.
17
18      C PACKAGE ROUTINES CALLED.. NONE HSPLVN 17
19      C USER ROUTINES CALLED.. NONE HSPLVN 18
20      C CALLED BY.. HSPLVN 19
21      C FORTRAN FUNCTIONS USED.. NONE HSPLVN 20
22
23      C
24      C DIMENSION X(1)*VNIKX)
25      C DIMENSION DELTAM(120)*DELTAP(120)
26      C DATA J/1*DELTAM/20*E-00+DELTAP/200.E-00/
27      C GO TO (10,20),INDEX
28
29      C
30      C J = 1
31      C VNIKX(1) = 1.
32      C IF (J .GE. JHIGH) GO TO 40
33      C 20 IPJ = ILFT+J
34      C DELTAP(J) = X(IPJ)-X
35      C IWPJ = ILFT-J+1
36      C DELTAM(J) = X - X(ILFT-J+1)
37      C VMPREV = 0.
38      C J+1 = J+1
39      C 10 30 L=1+J
40      C JPIML = JPI-L
41      C VM = VNIKX(L)/(DELTAP(L) + DELTAM(JPIML))
42      C VNIKX(L) = VM*DELTAP(L) + VMPREV
43      C VMPREV = VM*DELTAM(JPIML)
44      C VNIKX(JP1) = VMPREV
45      C J = JP1
46      C IF (J .LT. JHIGH) GO TO 20
47      C 40 PFJUPN
48      C FNU
  
```

CARD NR. SEVERITY DETAILS DIAGNOSIS OF PROBLEM
 21 1 AN IF STATEMENT MAY BE REPLICATED SO THAT A P ON A BRANCH (WHICH GOES TO STATEMENT

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76/76  OUT=1 MUNU=--*/ TACL   FIN 4.844H    04/15/80  11.10.37  PAGE: 4

1      SUBROUTINE INTEN ( X, LXT, X, ILEFT, MFLAG )           INTERV
2      C THIS SUBROUTINE IS PART OF THE H-SPLINE PACKAGE FOR THE STABLE          ??
3      C EVALUATION OF ANY H-SPLINE HAS THIS FUNCTION OR DERIVATIVE VALUE.        3
4      C See REFERENCE BELOW.          INTERV
5      C
6      C COMPUTES LARGEST ILEFT IN (X,LXT) SUCH THAT X(ILEFT) <= X. THE          INTERV
7      C PROGRAM STARTS THE SEARCH FOR ILEFT WITH THE VALUE OF ILEFT THAT WAS      4
8      C RETURNED AT THE PREVIOUS CALL (AND THIS SAVED IN THE LOCAL VARIABLE       INTERV
9      C IL0) TO MINIMIZE THE WORK IN THE CALL ON CASE THAT THE VALUE OF X ON     5
10     C THIS CALL IS CLOSE TO THE VALUE OF X ON THE PREVIOUS CALL. SHOULD      INTERV
11     C THIS ASSUMPTION NOT BE VALID, THEN THE PROGRAM LOCATES IL0 AND IH1      6
12     C SUCH THAT X(IL0) <= X < X(IH1) AND UNCE THEY ARE FOUND USES      INTERV
13     C BISECTION TO FIND THE CORRECT VALUE FOR ILEFT. MFLAG IS AN ERROR FLAG.  7
14     C FOR DEFINITIONS OF CALLING ARGUMENTS SEE ABOVE AND HSPLVD.          INTERV
15
16     C REFERENCE          INTERV
17
18     C DEMOHR, C., PACKAGE FOR CALCULATING WITH H-SPLINES. SIAM J.          INTERV
19     C NUMER. ANAL., VOL. 14, NO. 3, JUNE 1977. PP. 441-47.          19
20
21     C PACKAGE ROUTINES CALLED..  NONE          INTERV
22     C USER ROUTINES CALLED..  NONE          21
23     C CALLED BY..          COLPNT.LITAL VALUES          INTERV
24     C FORTRAN FUNCTIONS USED..  NONE          22
25
26
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SUBROUTINE INTERVAL 76/76 OPT=1 ROUND=--*/ TRACE FIN 4.R+4YH 04/15/H0 11.10.37 PA,t 32

```

    IF (X .LT. XT(IHI)) GO TO 50
    1STEP = 1STEP*2
    GO TO 41
40  IF (X .GT. XT(LX)) GO TO 110
    IHI = LX
    C NUM XT(IL0) .LE. X .LT. XT(IHI). RETURN THE INTERVAL.
    50 MIDDLE = (IL0 + IHI)/2
    IF (MIDDLE .EQ. IL0) GO TO 100
    C NOTE.. IT IS ASSUMED THAT MIDDLE = IL0 IN CASE IHI = IL0+1.
    C
    IF (X .LT. XT(MIDDLE)) GO TO 53
    IL0 = MIDDLE
    GO TO 50
53 IHI = MIDDLE
    GO TO 50
    C SET OUTPUT AND RETURN.
    90 MFLAG = -1
    ILEFT = 1
    RETURN
    100 MFLAG = 0
    ILEFT = IL0
    RETURN
    110 MFLAG = 1
    ILEFT = LX
    RETURN
    END
  
```

INTERV 59
 INTERV 60
 INTERV 61
 INTERV 62
 INTERV 63
 INTERV 64
 INTERV 65
 INTERV 66
 INTERV 67
 INTERV 68
 INTERV 69
 INTERV 70
 INTERV 71
 INTERV 72
 INTERV 73
 INTERV 74
 INTERV 75
 INTERV 76
 INTERV 77
 INTERV 78
 INTERV 79
 INTERV 80
 INTERV H1
 INTERV H2
 INTERV H3
 INTERV H4
 INTERV H5
 INTERV H6
 INTERV H7
 INTERV H8
 INTERV H9

SUBROUTINE STIF1H(Y,TOUT,YMAX,TH0R0,SAVE2,SAVE3)

P=IPV,WORK,WORK1)

C STIF1H PERFORMS ONE STEP OF THE INTEGRATION OF AN INITIAL VALUE
C PROBLEM FOR A SYSTEM OF ORDINARY DIFFERENTIAL EQUATIONS OF THE FORM
C $A(Y,T) \cdot (D(Y,T)) = G(Y,T)$, WHERE
C $Y = (Y(1), Y(2), \dots, Y(N))$.
C STIF1H IS FOR USE WHEN THE MATRICES "A" AND "G/DY" HAVE HANDED OR UNHANDLED FORM.
C THE DERIVATIVE OF $A(Y,T)$ ON Y IS ASSUMED TO BE ZERO.

C REFERENCE
C MINIMAH, A.C., PRELIMINARY DOCUMENTATION OF GEARIH. SOLUTION
C OF IMPLICIT SYSTEMS OF ORDINARY DIFFERENTIAL EQUATIONS WITH
C HANDLED JACOBIANS, LAWRENCE LIVERMORE LAB. UCID-30130, FEBRUARY
C 1976.

C COMMUNICATION WITH STIF1H IS DONE WITH THE FOLLOWING VARIABLES.
C Y AN NO BY LMAX ARRAY CONTAINING THE DEPENDENT VARIABLES
C AND THEIR SCALED DERIVATIVES. LMAX IS 13 FOR THE ADAMS
C METHODS AND 6 FOR THE BDF METHODS. LMAX - 1 = MAXHR
C IS THE MAXIMUM ORDER USED. SET SUBROUTINE CLOSE.
C Y(I,J+1) CONTAINS THE J-1TH DERIVATIVE OF Y(I). SCALD HY
C H**J/FACT0RIAL(J) (I = 0,1,...,N).
C NO A CONSTANT INTEGER. 0 GE. N. USED FOR DIMENSIONING PURPOSES.
C T THE INDEPENDENT VARIABLE. IT IS UPDATED ON EACH STEP TAKEN.
C H THE STEP SIZE TO BE ATTAINED ON THE NEXT STEP.
C H IS ALTERED BY THE ERROR CONTROL ALGORITHM DURING THE
C PROBLEM. H CAN BE EITHER POSITIVE OR NEGATIVE. BUT ITS
C SIGN MUST REMAIN CONSTANT THROUGHOUT THE PROBLEM.
C HMIN AND HMAX THE MINIMUM AND MAXIMUM ABSOLUTE VALUE OF THE STEP SIZE
C TO BE USED FOR THE STEP. THESE MAY BE CHANGED AT ANY
C TIME. BUT WILL NOT TAKE EFFECT UNTIL THE NEXT H CHANGE.
C EPS THE RELATIVE ERROR TOLERANCE. SEE DESCRIPTION IN PDFCOL.
C UPDOWN THE UNIT ROUNDOFF OF THE MACHINE.
C N THE NUMBER OF FIRST-ORDER DIFFERENTIAL EQUATIONS.
C MF THE METHOD FLAG. SEE DESCRIPTION IN PDFCOL.
C KFLAG A COMPLETION CODE WITH THE FOLLOWING MEANINGS.
C 0 THE STEP WAS SUCCESSFUL.
C -1 THE REQUESTED ERROR COULD NOT BE ACHIEVED
C WITH ABS(H) = HMIN.
C -2 THE REQUESTED ERROR IS SMALLER THAN CAN
C BE HANDLED FOR THIS PROBLEM.
C -3 CONVERGENCE COULD NOT BE
C ACHIEVED FOR ABS(H) = HMIN.
C -4 SINGULAR A-MATRIX ENCOUNTERED.
C ON A RETURN, WITH KFLAG NEGATIVE, THE VALUES OF I AND
C THF Y ARRAYS ARE AS OF THE BEGINNING OF THE LAST
C STEP. AND M IS THE LAST STEP SIZE ATTEMPTED.
C JSTART AN INTEGER USED ON INPUT AND OUTPUT.
C 0 INPUT, IT HAS THE FOLLOWING VALUES AND MEANINGS.
C 0 PERFORM THE FIRST STEP.
C 0 GT.0 TAKE A NEW STEP CONTINUING FROM THE LAST.
C -LT.0 TAKE THE NEXT STEP WITH A NEW VALUE OF
C H. EPS. NO. INDUCE 0.
C 0 EXIT. JSTART IS THE INDEX OF THE NEXT.
C YMAX AN ARRAY OF ELEMENTS WITH WHICH THE ESTIMATED LOCAL

```

76/76 OPT=1 MUND=-*/ I-ACT FIN 4.8.44H 04/15/80 11.10.37 PAHF 34

C ERRUR AN ARRAY OF N ELEMENTS. Y-WOM(1)/TU(2) IS THE ESTIMATED
C ONE-STEP ERRUR IN Y(1).
C SAVE1,SAVE2,SAVE3 THREE WORKING STORAGE ARRAYS, EACH OF LENGTH 14.
C PW A BLOCK OF LOCATIONS USED FOR THE CHORD ITERATION
C MATRIX. SEE DESCRIPTION IN PUTCOL.
C IPIV AN INTEGER ARRAY OF LENGTH N FOR PIVOT INFORMATION.
C ML,MU THE LOWER AND UPPER HALF MATRICES. SET RESPECTIVELY OF
C THE CHORD ITERATION MATRIX. SET DESCRIPTION IN PWCOL.
C WORK,WORK WORKING ARRAYS WHICH ARE USED TO PASS APPROPRIATE
C ARRAYS TO OTHER SUBROUTINES.
C PACKAGE ROUTINES CALLED.. COSET,DIFFUN,PSOLVE,PES,SOLN
C USER ROUTINES CALLED.. NONE
C CALLED BY.. PWCOL
C FURTHER FUNCTIONS USED.. AHS,AHAX,AMINI,FLUAT
C-
C DIMENSION Y(10,1),YMAX(N0),ERRUR(1),PW(1),IPIV(1),WOMK(1),SAVE2(N0),
C SAVE3(N0),WOM(1),IPIV(1),WOMK(1),WOMK(1)
C COMMON /SIZE3/ NINT,KOKO,NCC,NPH,NCP3,NEUN,IQUAD
C COMMON /START/ IWI(IW2,IW3,IW4,IW5,IW6,IW7,IW8,IW9,IW10,IW11),
C IW12,IW13,IW14,IW15,IW16,IW17,IW18
C COMMON /GEAR1/ T,M,HMIN,HMAX,EPS,HU,MU,MNU,MUM,NMF,KFLAG,JSTART
C COMMON/GEARP/EPS,HU,MU,MNU,MUM,NMF,KFLAG,JSTART
C COMMON/GEAR0/HUSEU,NUUSED,NUFL,NUFL,NFE,NFE
C COMMON/GEARSS/HMA
C COMMON/TARS/M,SMALL
C DIMENSION EL(13),TU(14)
C DATA EL(2)/1.0,0.0,0.0,0.0,0.0,0.0,0.0,0.0,0.0,0.0,0.0,0.0,0.0/
C KFLAG = 0
TOLD = T
IF (JSTART .GT. 0) GO TO 200
IF (JSTART .NE. 0) GO TO 120
C-
C ON THE FIRST CALL, THE ORDER IS SET TO 1, AND THE INITIAL Y0 IS
C CALCULATED. RMAX IS THE MAXIMUM VALUE BY WHICH M CAN BE INCREASED
C IN A SINGLE STEP. IT IS INITIALLY 1.E4 TO COMPENSATE FOR THE SMALL
C INITIAL M, BUT THEN IS NORMALLY EQUAL TO 10. IF A FAILURE
C OCCURS (IN CORRECTOR CONVERGENCE OR THROUGH TEST1, RMAX IS SET AT 2
C FOR THE NEXT INCREASE.
C-
N0 = 1
IEH = 0
CALL DIFFUN(INP,Y,SAVE1,IEH,M,IRIV,WOMK,WORK)
IF ( IEH .NE. 0 ) GO TO 645
NO 110 I = 10N
110 Y(1,2) = H*SAVE1(1)
METH = MF/10
MITER = MF = 10*M+1
L = 2
1100 HMAX = 3
HMAX = 1.E+04
XL = 0.
CHATE = 1.
PSOLD = FPS
PSOLD = FPS
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STIF1H 114
STIF1H 115

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SUBROUTINE STIF1H 7676 OPT=1 ROUND=--*/ (FACE
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115      MFOLD = MF
        NOLD = N
        NSTEP = 0
        NSTEPJ = 0
        NFT = 0
        NJE = 1
        INF = 3
        GO TO 130
120
125      C IF THE CALLER HAS CHANGED METH. CUST 1 IS CALLED TO SET
        C THE COEFFICIENTS OF THE METHOD. IF THE CALLER HAS CHANGED
        C N, EPS, OR METH, THE CONSTANTS E, EP, EUP, AND HNU MUST BE RESET.
        C E IS A COMPASSON FOR ERRORS OF THE CURRENT ODEUE NU. EP IS
        C TO TEST FOR INCREASING THE ODEUE. EP FOR DECREASING THE ODEUE.
        C AND IS USED TO TEST FOR CONVERGENCE OF THE CONNECTOR ITERATES.
        C IF THE CALLER HAS CHANGED M, Y MUST BE RESCALED.
        C IF HNU METH HAS BEEN CHANGED, IDUH.. IS RESET TO L + 1 TO PREVENT
        C FURTHER CHANGES IN H DUE THAT MANY STEPS.
120 IF (MF .EQ. MFOLD) GO TO 150
        MEU = METH
        METH = MF / 10
        MITER = MF - 10 * METH
        MFOLD = MF
        IF (MITER .NE. MIU) IMEVAL = MITER
        IF (METH .EQ. MEU) GO TO 150
        IDUH = L + 1
        IRET = 1
130 CALL COSET (METH, NU, EL, T4)
        LMAX = MAXER + 1
        HC = RC*EL(1)/OLDL0
        OLDL0 = EL(1)
140      FN = FLOAT(N)
        EUN = FN*(T0(1)*EPS)**2
        EL = FN*(T0(2)*EPS)**2
        EUP = FN*(T0(3)*EPS)**2
        HNU = FN*(T0(4)*EPS)**2
        GO TO 146, 170, 200, 201, 202
150      IF ((EPS .EQ. EPSOLD) .AND. (N .EQ. NOLD)) GO TO 160
        EPSOLD = EPS
        NOLD = N
        LKT = 1
        GO TO 140
160      IF (H .EQ. HOLD) GO TO 200
        HH = H/HOLD
        HOLD = 3
        GO TO 175
170      HH = AMAX1(HH,HMINV(AHS(H)))
175      HH = AMIN1(HH,HMAXV(AHS(H),HMAX))
        HH = AMIN1(HH,HMX/AHS(H))
        H1 = 1.
        DO 140 J = 2, L
        H1 = H1*HH
        DO 140 I = 1, J
        Y(I,J) = Y(I,J)*H1
170

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SUBROUTINE STIF1H 76/76 0PI= MOUND=-0/- TRACE FIN 4.8049H 04/15/80 11.10.37 PAGE 37

      * WORK*WUHK(1#14)*WUHK(1#15)* WUHK(1#16)*WUHK(1#17) *
      * WUHK(1#18)*K)
      C COMPUTE THE CORRECTOR ENUHK, & SUM WITH THAT AS RIGHT-HAND SIDE AND PW AS COEFFICIENT MATRIX.
      C USING THE LU DECOMPOSITION OF PW.
      C-----+
      230      CALL SOLINCPPTS(NL+ML+ML, P(1#1), SAV3(1#1), SAV3(1#2))
      DO 340 I=1,ST,LEN1
      ERHOK(1#1) = ERHOK(1#1) + SAVE3(1#1)
      U = U + (SAVE3(1#1)/YMAX(1#1))**2
      SAVE1(1#1) = Y(1#1) + EL(1#1)*ERHOK(1#1)
      340      SAVE2(1#1) = Y(1#1) + ERHOK(1#1)
      345      CONTINUE
      NFE=NFE+1
      C-----+
      C TEST FOR CONVERGENCE. If M.GT.0, AN ESTIMATE OF THE CONVERGENCE RATE CONSTANT IS STORED IN CRATE. A-4. THIS IS USED IN THE TEST.
      C-----+
      440 IF (M .NE. 0) CRATE = AMAX1(1.9*(UMATE*U/U1),
      C IF (ID*AMIN11.0E-2,*CRATE)).LE.*IND GO TO 450
      U1 = U
      M = M + 1
      IF (M .EQ. 3) GO TO 410
      GO TO 360
      C-----+
      C THE CORRECTOR ITERATION FAILED TO CONVERGE IN 3 ITER.
      C THE Y ARRAY IS REHEATED TO ITS VALUES
      C BEFORE PREDICTION, AND H IS REDUCED, IF POSSIBLE. IF NOT, A
      C NO-CONVERGENCE EXIT IS TAKEN.
      C-----+
      455      CONTINUE
      460      I=VAL=MIFR
      460      T = TOLD
      RMAX = 2.
      DO 430 J1 = 1,NQ
      DO 430 J2 = J1,NQ
      J = (NQ + J1) - J2
      DO 430 I = 1,N
      Y(I,J) = Y(I,J) - Y(I,J-1)
      430      IF (AHS(H) .LE. HMIN*1.00001) GO TO 640
      H=0.60
      TENDO = 1
      GO TO 170
      C-----+
      C THE CORRECTOR HAS CONVERGED. IWEAL IS SET TO -1 TO SIGNAL
      C THAT PW MAY NEED UPDATING ON SUBSEQUENT STEPS. THE ERROR TEST
      C IS MADE AND CONTROL PASSES TO STATEMENT 500 IF IT FAILS.
      C-----+
      475      IWEAL = -1
      O = 0.
      DO 460 I = 1,N
      460      O = O + (ERHOK(1#1)/YMAX(1#1))**2
      IF (O .GT. E) GO TO 500
      C-----+
      C AFTER A SUCCESSFUL STEP, UPDATE THE I AKAY.
      C CONSIDER CHANGING H IF IWEAL = 1. (THE FIRST WHETHER LAST)
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SUBROUTINE STIFH      76/76   OPT=1 KOUND=-++/ TRACE          FIN 4.8+498
04/15/80  11.10.37  PAGE    38

C IF IDOUB IS THEN 1 AND NO .LT. MAXH, THEN EMKOR IS SAVED FOR
C USE IN A POSSIBLE ORDER INCREASE IN THE NEXT STEP.
C IF A CHANGE IN H IS CONSIDERED, AN INCREASE OR DECREASE IN ORDER
C BY ONE IS CONSIDERED ALSO. A CHANGE IN H IS MADE ONLY IF 11 IS BY A
CC FACTOR OF AT LEAST 1.1. IF NOT, LUMH IS SET TO 3 TO PREVENT
C TESTING FOR THAT MANY STEPS.
C H IS NOT INCREASED NEAR TOUT.
C--          STIFH 287
290          STIFH 288
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          STIFH 342
          STIFH 343

470 KFLAG = 0          STIFH 287
        IWE0 = 0          STIFH 288
        NSTEP = NSTEP + 1  STIFH 289
        HUSED = H          STIFH 290
        NUSED = NO         STIFH 291
        DO 470 J = 1,L    STIFH 292
          DO 470 I = 1,N  STIFH 293
            Y(I,J) = Y(I,J) + E(I,J)*EMKOR(I)  STIFH 294
            IF (I+2.0*M.GT.TOUT) IDOUB=IDOUB+1  STIFH 295
            IDOUB=IDOUB*M
            IF (IDOUB .EQ. 1) GO TO 520  STIFH 296
            IDOUB = IDOUB - 1  STIFH 297
            IF (IDOUB .GT. 1) GO TO 700  STIFH 298
            IF (L .EQ. LMAX) GO TO 700  STIFH 299
            DO 490 I = 1,N  STIFH 300
              Y(I,LMAX) = ERROR(I)  STIFH 301
              GO TO 700  STIFH 302
C--          STIFH 303
C THE ERROR TEST FAILED. KFLAG KEEPS TRACK OF MULTIPLE FAILURES.
C RESTORE T AND THE Y ARRAY TO THEIR PREVIOUS VALUES, AND PREPARE
C TO TRY THE STEP AGAIN. COMPUTE THE OPTIMUM STEP SIZE FOR THIS ORDER.
C--          STIFH 304
310          KMAX = KFLAG - 1  STIFH 305
          T = TOLD  STIFH 306
          DO 510 J1 = 1,N  STIFH 307
            DO 510 J2 = J1,N  STIFH 308
              J = (NO + J1) - J2  STIFH 309
              DO 510 I = 1,N  STIFH 310
                Y(I,J) = Y(I,J) - Y(I,J+1)  STIFH 311
C--          STIFH 312
315          KMAX = 2.  STIFH 313
          IF (AHS(H) *LE. MIN(1.00001) ).. TO 660  STIFH 314
          IF (KFLAG .LE. -3) GO TO 640  STIFH 315
          IMDO = 2  STIFH 316
          ENQ2=5.0/FLOAT(L)  STIFH 317
          PR2=(L/E)*ENQ2*2.0+2.0E-06  STIFH 318
          HH=1.0/PR2  STIFH 319
          GO TO 170  STIFH 320
C--          STIFH 321
320          PH1=PH2  STIFH 322
          PH2=PH3  STIFH 323
          ENQ2=5.0/FLOAT(L)  STIFH 324
          PR2=(L/E)*ENQ2*2.0+2.0E-06  STIFH 325
          HH=1.0/PR2  STIFH 326
          GO TO 170  STIFH 327
C REGARDLESS OF THE SUCCESS OR FAILURE OF THE STEP, FACTORS
C PH1, PH2, AND PH3 ARE COMPUTED. H WHICH H COULD BE DIVIDED
C AT ORDER NO - 1. ORDER NO. 0H WHICH NO + 1, RESPECTIVELY.
C IN THE CASE OF FAILURE, PH3 = 1.E20 TO AVOID AN ORDER INCREASE.
C THE SMALLEST OF THESE IS DETERMINED AND THE NEW ORDER CHOSEN;
C ACCORDINGLY, IF THE ORDER IS TO BE INCREASED, WE COMPUTE OUR
C ADDITIONAL SCALED DERIVATIVE.
C--          STIFH 328
325          ENQ3 = 1.E+20  STIFH 329
          IF (L .EQ. LMAX) GO TO 540  STIFH 330
          ENQ3 = 1.E+20  STIFH 331
          ENQ3 = 1.E+20  STIFH 332
          ENQ3 = 1.E+20  STIFH 333
          ENQ3 = 1.E+20  STIFH 334
          ENQ3 = 1.E+20  STIFH 335
          ENQ3 = 1.E+20  STIFH 336
          ENQ3 = 1.E+20  STIFH 337
          ENQ3 = 1.E+20  STIFH 338
          ENQ3 = 1.E+20  STIFH 339
          ENQ3 = 1.E+20  STIFH 340
          ENQ3 = 1.E+20  STIFH 341
          ENQ3 = 1.E+20  STIFH 342
          ENQ3 = 1.E+20  STIFH 343

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SUBROUTINE STIF1H 76/76 OPT=1 MOUND=-*/ TRACE FTN 4.84498 04/15/80 11.10.J7 PAGE 40

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400      IF ( IER .NE. 0 ) GO TO 685
        NJE = NJE + 1
        DO 650 I = 1,N
650      Y(I,2) = H*SAVE1(I)
        IVAL = MITER
        IDOUR=3
        IF (INV .EQ. 1) GO TO 200
        INV = 1
        L = 2
        INT1 = 3
        GO TO 130
C ALL RETURNS ARE MADE THROUGH THIS SECTION. H IS SAVED IN HOLD
C TO ALLOW THE CALLER TO CHANGE H ON THE NEXT STEP.
130
410
415      660 KFLAG = -1
        GO TO 700
670      KFLAG = -2
        GO TO 700
        680 KFLAG = -3
        GO TO 700
685      KFLAG = -4
        GO TO 700
        690      HMAX=10.0
        700      MULD = H
        START = NO
        C WE DO NOT WANT ANY NEGATIVE CONCENTRATIONS.
        RETURN
        END
      
```

CARD NR. SEVENTIY DETAILS DIAGNOSIS OF PROBLEM
 153 1 AN IF STATEMENT MAY BE MORE EFFICIENT THAN A 2 OR 3 BRANCH COMPUTED GO TO STATEMENT.

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SUBROUTINE GFUN 76/76  OPT=1 MOUND=++/ TRACE   FIN 4.0.H-44H  04/15/90  11.10.37  PAGE 41

1      SUBROUTINE GFUN (IT,LC,UDF,NPDE, I PTS,A,DC,CDR,U,UMR,K,IL,I,J)          GF IN    ?
*     AC(UVAL,IL,FT,KNP,SKT,KNS)           GF IN    3
*     CALLING ARGUMENTS ARE DEFINED BELOW AND IN PROLOG.          GF IN    4
*-----*
5      C SUBROUTINE GFUN COMPUTES THE FUNCTION, UDEF(X(0,1)), THE WEIGHT-          GF IN    5
*     AND STATE OF THE SEMI-DISCRETE APPROXIMATION TO THE ORIGINAL          GF IN    6
*     SYSTEM OF PARTIAL DIFFERENTIAL EQUATIONS AND UPDATES THE BOUNDARY          GF IN    7
*     CONDITION INFORMATION.          GF IN    8
*-----*
10     C PACKAGE ROUTINES CALLED...          EVAL          GF IN    9
*     USEW MULTIMIN'S CALLED...          HMDRY,OF          GF IN    10
*     CALLED BY...          DIFFUN,PAR,THRETS          GF IN    11
*     FINTWAN FUNCTIONS USED...          NONE          GF IN    12
*-----*
15     DIMENSION C(1),UDF(1)          GF IN    13
      DIMENSION A(1),HC(INPUT,4),XC(1),UVAL(INPUT,3),ILEF(1)          GF IN    14
      DIMENSION DZT(INPUT),DRNU(INPUT),DRNU(INPUT),IQUAD          GF IN    15
      COMMON /SIZES/ NINT,NUND,IUND,IQUAD          GF IN    16
      DO 10 J=1,4          GF IN    17
      MC(KPDE+J)=0.0          GF IN    18
*-----*
20     C UPDATE THE LEFT BOUNDARY VALUES.  SAVE LEFT BOUNDARY CONDITION          GF IN    19
*     CC INFORMATION IN THE FIRST 2NPDE LOCATIONS OF MC.          GF IN    20
*-----*
25     C NDF... UVAL(K,1) = U(K) * UVAL(K,2) = U(K) * AND UVAL(K,3) = U(K).          GF IN    21
*-----*
30     KS=(NPDE-1)*NPTS+1          GF IN    22
      CALL EVAL(1,NPDE,C,ILVAL,A,ILFT)          GF IN    23
      CALL FT(XC(1),UVAL(1,2),UVAL(1,3),UVAL(1,4),KS)          GF IN    24
      * NPDE,NPDE+1,KS1,KS2          GF IN    25
      CALL HMDRY(C,XC(1),UVAL,UVAL(1,2),UVAL,UVAL(1,3),UVAL,UVAL(1,4))          GF IN    26
      ILIM = KNP + 2          GF IN    27
      MC(KPDE+1)=1.0          GF IN    28
      IF (DRNU(KPDE+1)=0.0 .AND. DRNU(KPDE+2)=0.0) GO TO 30          GF IN    29
      IUNIT(KS)=DZT(KPDE+1)          GF IN    30
      MC(KPDE+2)=A(ILIM)+DRNU(KPDE+1)          GF IN    31
      MC(KPDE+3)=A(ILIM)+DRNU(KPDE+2)          GF IN    32
      30  CONTINUE          GF IN    33
*-----*
35     C MAIN LOOP TO FORM WEIGHT SITE OF ONE AT THE COLLOCATION POINTS.          GF IN    34
*-----*
40     ILIM = NPTS - 1          GF IN    35
      DO 40 I=2,ILIM          GF IN    36
      CALL EVAL(1,NPDE,C,UVAL,A,ILFT)          GF IN    37
      NS=KS1          GF IN    38
      CALL FT(XC(1),UVAL,UVAL(1,2),UVAL(1,3),UVAL(1,4),NS)          GF IN    39
      * NPDE,NPDE+1,KS1,KS2          GF IN    40
      40  CONTINUE          GF IN    41
*-----*
45     C INITIATE THE WEIGHT BOUNDARY VALUES.  SET THE WEIGHT BOUNDARY CONDITIONS          GF IN    42
*     CC INFORMATION IN THE LAST 2NPDE LOCATIONS OF MC.          GF IN    43
*-----*
50     C ALL EVAL(NPDE,...,C,UVAL,A,ILFT)          GF IN    44
      * NPDE,NPDE+1,KS1,KS2          GF IN    45
      C ALL EVAL(NPDE,...,C,UVAL,A,ILFT)          GF IN    46
      * NPDE,NPDE+1,KS1,KS2          GF IN    47
      C ALL EVAL(NPDE,...,C,UVAL,A,ILFT)          GF IN    48
      * NPDE,NPDE+1,KS1,KS2          GF IN    49
      C ALL EVAL(NPDE,...,C,UVAL,A,ILFT)          GF IN    50
      * NPDE,NPDE+1,KS1,KS2          GF IN    51

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SUMMOUT INT	GFUN	76/76	OPT=1	MOUND=--*/	1-ACT	FIN 4.0.0495	04/15/H0	11.10.37	PAC.E	4?	
60						CALL MNDY(1.T,XC(MCPT5).UVAL,UVAI (1.02).0.0HJU0.0DZD1.NPUE) ILIM = MCPTS * J * KURD - KOM - 1 HC (KPOE,4)=1.0 IF (DNDU(KPDE).EU.0.0.AND.DNDU(A(KPDE)).EU.0.0)GO TO 60 UDUT(KS)=DZD1(KPDE) HC (KPOE,3)=A(ILIM).DNDU(A(KPDE)) HC (KPOE,4)=DNDU(KPDE) +IC(KPDE+3) 60 COUNTINUE HT TURN END					
61							GF UN	59			
62							GF UN	60			
63							GF UN	61			
64							GF UN	62			
65							GF UN	63			
66							GF UN	64			
67							GF UN	65			
68							GF UN	66			

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SUBROUTINE EVAL
    76/76  OPT=1 MUNU=-#/- (~ACT
          F7N 4.444H
          06/15/H0 11.10.37
          PAGE 43

1      SUBROUTINE EVAL(IICPT,NPDE,C,UVAL,A,ILEFT)
          EVAL 2
          EVAL 3
          EVAL 4
          EVAL 5
          EVAL 6
          EVAL 7
          EVAL 8
          EVAL 9
          EVAL 10
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          EVAL 28
          EVAL 29
          EVAL 30

C----- CALLING ARGUMENTS ARE DEFINED BELOW AND IN WHICUL.
C----- CALLING ARGUMENTS ARE DEFINED BELOW AND IN WHICUL.

5      C SUBROUTINE EVAL EVALUATES U(K) . AND U(A(K)) . K=1 TO NPDE .
      C AT THE COLLOCATION POINT WITH INDEX IICPT USING THE VALUES OF
      C THE BASIS FUNCTION COEFFICIENTS IN C AND THE BASIS FUNCTION VALUES
      C STORED IN A. THE RESULTS ARE STORED IN UVAL AS FOLLOWS.
      C UVAL(K,1) = U(K) . UVAL(K,2) = U(A(K)) . AND UVAL(K,3) = UXX(K).
      C
10     C PACKAGE ROUTINES CALLED..   NONE
      C USEW ROUTINES CALLED..   NONE
      C CALLED HY..   GFUN,PUEUL,PSERT1H
      C FORTAN FUNCTIONS USED..   NONE
      C
15     DIMENSION C(1) UVAL(NPDE+3)*A(1)*ILEFT(1)
      COMMON/SIZE/NINT,KORD,IICPT(12)*,OPTS,ONE,D1WAD
      IK = ILEFT(IICPT) - KORD
      IC = 3*KORD*IICPT-1
      DO 10 M=1,3
      ICC = IC + KOM*(M-1)
      DO 10 J=1,NPDE
      JC=(J-1)*NCPTS
      UVAL(J,M) = 0.
      DO 10 I=1,KORD
      UVAL(I,J,M)=UVAL(J,M)+C(JC+I+IK)*A(I*IICPT)
      10 CONTINUE
      RETURN
      END

```

```

SUBROUTINE DIFFUN 76/76  OPT=1 MOUNDU==*/ IMACT   F7N 4.8.49H    04/15/80  11.10.37    PAGE: 44

1      SUBROUTINE DIFFUN (N, Y, YDU), IER, PW, IPIV, WORK, IWORK      DIFFUN    2
C-----CALLING ARGUMENTS ARE DEFINED BELOW AND IN PDTCOL.
C-----THIS ROUTINE COMPUTES YDU = A(Y,T)**-1 * G(Y,T) BY USE OF
C-----THE ROUTINES GFUN, ADUA, DECD, AND SOLH.
C-----PACKAGE ROUTINES CALLED-- ADUA,DECD,GFUN,SOLH      DIFFUN    3
C-----USER ROUTINES CALLED-- NONE      DIFFUN    4
C-----CALFD HY.      DIFFUN    5
C-----FURTHER FUNCTIONS USED-- STIFIH      DIFFUN    6
C-----COMMON /GEAR9/ EPSJ,HU,ML,MU,MN,NOML,NUM,NUM1      DIFFUN    7
C-----COMMON /SIZES/ NINT,KORD,NCC,NP,NCPTS,NCNT,QUAD      DIFFUN    8
C-----COMMON /START/ IWI,IW2,IW3,IW4,IW5,IW6,IW7,IW8,IW9,IW10,IW11,IW12,IW13,IW14      DIFFUN    9
C-----DIMENSION Y(N),YDOT(N),PW(1),IPIV(1),WORK(1),IWORK(1)      DIFFUN    10
C-----DIFFUN    11
C-----DIFFUN    12
C-----DIFFUN    13
C-----DIFFUN    14
C-----DIFFUN    15
C-----DIFFUN    16
C-----DIFFUN    17
C-----DIFFUN    18
C-----DO 10 I = 1,NOW      DIFFUN    19
C-----PW(I) = 0.      DIFFUN    20
C-----DO 20 K=1,NPDE      DIFFUN    21
C-----CALL GFUNIT(Y,YDU),NPDE,NCPTS,NINT,K,WORK,IWORK,K      DIFFUN    22
C-----WORK(IW15)=WORK(IW16),WORK(IW17)=IWORK(K),WORK(K)      DIFFUN    23
C-----NST=(K-1)*NQ1+1      DIFFUN    24
C-----NST=(K-1)*NCPTS+1      DIFFUN    25
C-----CALL ADDA(PW(NST)),NCPTS,WORK(IW1),IWORK,NPDE,K      DIFFUN    26
C-----CALL DECB(NCPTS,NCPTS,ML,MU,P,IWT),IPIV(IWT),IEN      DIFFUN    27
C-----IF (IER,NE,0) RETURN      DIFFUN    28
C-----CALL SOLR(NCPTS,NCPTS,ML,MU,P,W(NST),YDUT(IWT),IPIV(IWT))      DIFFUN    29
C-----CONTINUE      DIFFUN    30
C-----RETURN      DIFFUN    31
C-----END      DIFFUN    32

```

SUBROUTINE ADDA 76/76 OPT=1 KINUND=-/- I-ACT F7N 4.044H

```
1      SUBROUTINE ADDA(PW,IND,A,ILEFT,BL,NUUE,KPDE)
2
3      C-----C CALLING ARGUMENTS ARE UNDEFINED BELOW AND IN PICTOL AND STIFTH.
4
5      C-----C SUBROUTINE ADDA ADDS THE MATRIX A TO THE MATRIX STORED IN PW IN
6      C-----C HAND FORM. PW IS STORED BY DIAGONALS WITH THE LUMEMOST DIAGONAL
7      C-----C STORED IN THE FIRST COLUMN OF THE ARRAY.
8      C-----C ONLY THE PART OF PW CORRESPONDING TO THE KPUT OF PW IS
9      C-----C WORKED WITH.
10     C-----C PACKAGE ROUTINES CALLED.. NONE
11    C-----C USER ROUTINES CALLED.. NONE
12    C-----C CALLED BY.. DIFFUR,PSRTIM
13    C-----C FORTRAN FUNCTIONS USED.. NONE
14
15    DIMENSION PW(IND+1),A(1),ILEFT(1),HC INPUT(4),
16    COMMON /SIZES/ NINT,NINT,KORD,NCC,NPI,NCPTS,NEUN,LUAD
17
18    C-----C ADD THE BOUNDARY CONDITION PORTIONS OF THE A MATRIX TO PW ( THE FIRST
19    C-----C AND LAST ROWS).
20
21    ICOL=ILEFT(1)+LUAD-1
22    PW(1,ICOL)=PW(1,ICOL)+AC(KPDE+1)
23    ADUA
24    PW(1,ICOL+1)=PW(1,ICOL+1)+AC(KPDE+2)
25    PW(NCPTS,ICOL-1)=PW(NCPTS,ICOL-1)+HC(KPDE+3)
26    PW(NCPTS,ICOL)=PW(NCPTS,ICOL)+HC(KPDE+4)
27
28    C-----C UPDATE THE REMAINING ROWS OF PW BY ADDING THE APPROPRIATE VALUE'S
29    C-----C IN A TO PW.
30
31    IND = NCPTS - 1
32    DO 20 I=2,IND
33    I2 = (I-1)*KORD + 3
34    ICOL = ILEFT(I) - 1 + LUAD - 1
35
36    DO 20 J=1,KORD
37    JI=ICOL+J
38    J2 = I2 + J
39    PW(I,J1)=PW(I,J1)+A(J2)
40    IF TURN
41
42
```

SUBROUTINE RES 76/76 0*T=1 HROUND=--*/ 1-ACE FIN 4.8+498 04/15/80 11.10.37 PAGE 46
 * XC.UVAL.K1
 C CALLING ARGUMENTS ARE DEFINED RELYIN AND IN P1COL.
 C SUBROUTINE RES COMPUTES THE RESIDUAL VECTOR $H = H^*G(C,T) - A(C,T)*V$
 C WHERE H IS THE CURRENT TIME STEP S1ST. G IS A VECTOR. A IS A
 C MATRIX. V IS A VECTOR. AND T IS THE CURRENT TIME.
 CC ONLY THE PART OF H CORRESPONDING TO THE KTH PUE IS CALCULATED
 CC IN A SINGLE CALL TO RES.
 C PACKAGE ROUTINES CALLED.. GFUN
 C USER ROUTINES CALLED.. NONE
 C CALLED BY.. STIFH
 C FUNKTRAN FUNCTIONS USED.. NONE
 C
 DIMENSION C(1),R(1),V(1)
 DIMENSION A(1),ILEFT(1),BC(1),XC(1),UVAL(1)
 DIMENSION DHDUX(INPUT),DBDUX(INPUT),DZDT(INPUT)
 COMMON /SIZE/ NINT,KORD,NCC,1D1,(3),IQUAD
 C FORM G(C,T) AND STORE IN H.
 CALL GUNIT,C,R,INPUT,NCPTS,A,BL,UHDUX,DZDT,XC,IWAL,
 * ILEFT,K,KOK
 C FORM THE FIRST AND LAST BLOCK ROWS OF THE RESIDUAL VECTOR
 C WHICH ARE DEPENDENT ON THE BOUNDARY CONDITIONS.
 ILIM = NCPTS - 1
 I1ST = (K-1)*NCPTS + 1
 IEND = K*NCPTS
 SUM1 = HC(K,1)*V(I1ST)*AC(K,2)*V(I1ST+1)
 SUM2 = BC(K,3)*V(IEND-1)*BC(K,4)*V(IEND)
 H(I1ST) = H(I1ST) - SUM1
 H(IEND) = H(IEND) - SUM2
 C FORM THE REMAINING COMPONENTS OF THE RESIDUAL VECTOR.
 C
 DO 50 ICPTS=2,ILIM
 ICOL = (ICPTS-1)*KORD + 3
 SUM1 = 0.
 DO 30 J=1,KORD
 SUM1 = SUM1 + A((J2+J)*V(I1ST-1+ICOL)+J)
 JU
 H(I1ST-1+ICPTS) = H(I1ST-1+ICPTS) - SUM1
 40 CONTINUE
 50 CONTINUE
 RETURN
 END


```

SUBROUTINE PSETIH 76/76  OPT=1 MOUND=--*/ 1-ACE   FTN 4.0.H49H  04/15/80 11.10.37
      PSETIH 59
      PSETIB 60
      PSETIH 61
      PSETIH 62
      PSETIH 63
      PSETIB 64
      PSETIB 65
      PSETIH 66
      PSETIH 67
      PSETIH 68
      PSETIH 69
      PSETIH 70
      PSETIH 71
      PSETIB 72
      PSETIH 73
      PSETIB 74
      PSETIB 75
      PSETIB 76
      PSETIH 77
      PSETIB 78
      PSETIB 79
      PSETIH 80
      PSETIB 81
      PSETIH 82
      PSETIB 83
      PSETIB 84
      PSETIB 85
      PSETIB 86
      PSETIB 87
      PSETIB 88
      PSETIB 89
      PSETIB 90
      PSETIB 91
      PSETIB 92
      PSETIB 93
      PSETIH 94
      PSETIB 95
      PSETIB 96
      PSETIB 97
      PSETIH 98
      PSETIH 99
      PSETIB 100
      PSETIB 101
      PSETIH 102
      PSETIB 103
      PSETIH 104
      PSETIB 105
      PSETIH 106
      PSETIB 107

      NEND=K*NOH1
      U0 30 KHLK=KL1*U0*KUP
      J1=ICOL*KBLK
      J2 = 12 * KHLK
      J3 = J2 + KOMU
      J4 = J3 + KOMU
      PW(INST-1)+(J1-1)*NCPTS1=DFUU(K)*A(J2)*UFUU(X(K))*A(J3)
      *+DFUU(X(K))*A(J4)

      30 CONTINUE
      C MODIFY THE LAST AND THE FIRST HLUCK ROWS FOR THE BOUNDARY CONDITIONS.
      C CURRENT INFORMATION FOR THE RIGHT BOUNDARY CONDITION IS ALREADY IN
      C THE ARRAYS DROW, DRDX AS A RESULT OF A PREVIOUS CALL TO GFUN.
      C

      DU 50 K=1,NPUE
      NST=(K-1)*NOH1+1
      NEND=K*NOH1
      IF (DRDU(K),EQ.0.0.ANU.DHUU(X(K),T)=0.0.01GU 10 50
      DO 40 J=1,MW
      PW(INST-1)+J*NCPTS1=0.0
      40  CONTINUE
      50 CONTINUE
      CALL EVAL(1,NPDE,C,UVAL,A,ILEFT)
      CALL HNDY(T,XC(1)),UVAL,UVAL(1,2),DHUU,UHUU,X(DZUT,NPDE)
      DU 70 K=1,NPDE
      NST=(K-1)*NOH1+1
      NEND=K*NOH1
      IF (DRDU(K),EQ.0.0.ANU.DBDDUX(K)*EU,0.0)GU 10 70
      DU 60 J=1,MW
      PW(INST+(J-1)*NCPTS1)=0.0
      60  CONTINUE
      70 CONTINUE
      DU R0 T=1,NOH
      PW(1)=PW(1)*CON
      C AND MATRIX A(C,T) TU PW.
      C

      DU 90 K=1,NPUE
      NST=(K-1)*NOH1+1
      NEND=K*NOH1
      IST=(K-1)*NCPTS1+1
      CALL ADDA(IP(INST),NCPTS1,A,ILEFT,T,C,NPDE,K)
      C DU LU DECOMPOSITION IN PW.
      C CALL UCH(NCPTS1,NCPTS1,MU,PW(1,IST),IPV(1,IST),JtK)
      90  CONTINUE
      END

```

```

1      SUBROUTINE DIFF (T,X,IPT,UUX,UAA,UDU,UDLUX,UDUXX,NPU,CMAX)
2      *          SAVE2)
3
4      C-----C CALLING ARGUMENTS ARE DEFINED BELOW AND IN MULUL.
5      C-----C SUBROUTINE DIFF IS USED IF MITCH=2 TO PROVIDE FINITE DIFFERENCE
6      C APPROXIMATIONS FOR THE PARTIAL DERIVATIVES OF THE K-TH USER DEFINED
7      C FUNCTION IN THE F ROUTINE WITH RESPECT TO THE VARIABLES U, UX, AND
8      C UXX. THESE PARTIALS WITH RESPECT TO U, UX, AND UXX ARE COMPUTED,
9      C STORED, AND RETURNED IN THE INPUT VECTORS UDU,UDUX, AND UDUXX
10     C RESPECTIVELY AT CALCULATION POINT NUMBER IPI. AGAIN, PARTIALS
11     C OF THE KTH FUNCTION WITH RESPECT TO THE JTH VARIABLE, K NOT
12     C EQUAL TO J, ARE ASSUMED TO BE NEGLIGIBLE.
13
14     C PACKAGE ROUTINES CALLED...   NONE
15     C USEK Routines CALLED...    F
16     C CALLED HY...             PSETIH
17     C FURTHER FUNCTIONS USED... AMAX1
18
19     C-----C DIMENSION U(NPDE),UA(NPDE),UXX((NPDE)*UDU(NPDE))
20     C-----C DIMENSION UDU(X(NPDE)),UDUX(NPDE),UDUXX(NPDE),
21     C-----C COMMON/GEAR9/EPSJ,RHOML,NUMM,NML,NDML,NUMN,NUNI
22     C-----C COMMON/SIZES/NINT,KUND,NCC,NPU,PCPTS,NEUN,IUAU
23     C-----C KSKR=2
24     C-----C KSKT=2
25     C-----C DU 40 K=1,NPDE
26     C-----C IF (K.EQ.NPDE) KSKH=1
27     C-----C I1=(K-1)*NCPTS+1PT
28     C-----C UK=U(I1)
29     C-----C H=EPSJ*CMAX(I1)
30     C-----C W = AMAX1(R,R0)
31     C-----C U(K)=U(I1)*R
32     C-----C KINV = 1. / R
33     C-----C CALL F (T,X,U,UX,UXX,UDU (K)*NP1,E,K,IPT+1,KSKH)
34     C-----C DFDU(K)=(DFDU(K)-SAVE2(I1))/KINV
35     C-----C U(K)=UK
36     C-----C UK=UX(I1)
37     C-----C U(K)=UX(I1)+W
38     C-----C CALL F (T,X,U,UX,UXX,UDUX (K)*NP1,E,K,IPT+1,KSKH)
39     C-----C DFLUX (K)=(DFLUX(K)-SAVE2(I1))/KINV
40     C-----C UK (K)=UK
41     C-----C UK=UX (K)
42     C-----C U(K)=UX (K)+W
43     C-----C CALL F (T,X,U,UX,UXX,UDUX (K)*NP1,E,K,IPT+1,KSKH)
44     C-----C DFLUX (K)=(DFLUX(K)-SAVE2(I1))/KINV
45     C-----C UK (K)=UK
46     C-----C UK=UX (K)
47     C-----C CONTINUE
48     C-----C RETURN
49

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SUBROUTINE INTMP      76/76   0PF=1 ROUND=---/ TRACT      F FN 4.0.H+44H
          06/15/H0 11.10.37      PAGE  50

1      SUBROUTINE INTMP (TOUT, Y, NU, T0)
C CALLING ARGUMENTS ARE DEFINED BELOW ...IN STMT 14
C SUBROUTINE INTMP COMPUTES INTERPOLATE VALUES OF THE DEPENDENT
C VARIABLE Y AND STORES THEM IN Y0.  THE INTERPOLATION IS TO THE
C POINT T = TOUT.  AND USES THE MODULUS K HISTORY ARRAY Y, AS FOLLOWS...
C
C          Y0(I) = SUM V(I,J+1)*S(J) .
C          WHERE S = -(T-TOUT)/H.
C          USES MUNITES CALLED...    NONE
C          USER MUNITES CALLED...  NONE
C          CALLED HY...           HDECUL
C          FORTNAN FUNCTIONS USED...  NONE
C
C----- DIMENSION Y(NU), Y(NU,1)
C----- COMMON /GEAR1/ T,H,DUMMY(14),NU,J,UMMY(2),JSTART
C----- DO 10 I = 1,N
10     Y0(I) = V(I,1)
        L = JSTART + 1
        S = (TOUT - T)/H
        SI = 1,
        LU 30 J = 2*L
        SI = SI+S
        LU 20 I = 1+N
        20   Y0(I) = Y0(I) + S*Y(I,J)
        30 CONTINUE
        RETURN
      END
C----- 2
C----- 3
C----- 4
C----- 5
C----- 6
C----- 7
C----- 8
C----- 9
C----- 10
C----- 11
C----- 12
C----- 13
C----- 14
C----- 15
C----- 16
C----- 17
C----- 18
C----- 19
C----- 20
C----- 21
C----- 22
C----- 23
C----- 24
C----- 25
C----- 26
C----- 27
C----- 28
C----- 29
C----- 30
C----- 31
C----- 32

```

SUBROUTINE COSET (METH, NJO, EL, 10)

C 10 COSET IS CALLED BY THE INTEGRATOR AND SETS COEFFICIENTS USED IN THE
 C 11 THE VECTOR EL OF LENGTH NO + 1. DETERMINES THE BASIC METHOD.
 C 12 THE VECTOR EL OF LENGTH NO + 1. DETERMINES THE STEP SIZE.
 C 13 THE VECTOR TO, OF LENGTH 4, IS INVOLVED IN ADJUSTING THE STEP SIZE
 C 14 IN RELATION TO TRUNCATION ERROR. ITS VALUES ARE GIVEN BY THE
 C 15 PERTST ARRAY.
 C 16 THE VECTORS EL AND TO DEPEND ON METH AND NO.
 C 17 THE MAXIMUM ORDER, MAXORD, OF THE METHODS AVAILABLE IS CURRENTLY
 C 18 12 FOR THE ADAMS METHODS AND 5 FOR THE RKF METHODS. MAXORD DEFAULTS
 C 19 TO 5 UNLESS THE USER SETS MAXORD TO SOME OTHER LEGITIMATE VALUE.
 C 20 THROUGH THE COMMON BLOCK /OPTION/. SEE PDECUL FOR ADDITIONAL DETAILS.
 C 21 IMAX = MAXORD + 1 IS THE NUMBER OF COLUMNS IN THE Y ARRAY (SEE S111H)
 C 22 AND THE VARIABLE CO_Y UN WORK(111) IN PDECUL.
 C 23
 C 24 THE COEFFICIENTS IN PERST NEED NOT BE GIVEN TO UNILY ABOUT
 C 25 ONE PERCENT ACCURACY. THE ORDER IN WHICH THE GROUPS APPEAR BELOW
 C 26 IS COEFFICIENTS FOR ORDER NO - 1, COEFFICIENTS FOR ORDER NO.
 C 27 COEFFICIENTS FOR ORDER NO + 1, WITHIN EACH GROUP AND THE
 C 28 COEFFICIENTS FOR THE ADAMS METHODS FOLLOWED BY THOSE FOR THE
 C 29 RKF METHODS.
 C 30
 C 31 REFERENCE
 C 32 GEAR, C.W., NUMERICAL INITIAL VALUE PROBLEMS IN OKINAWA
 C 33 DIFFERENTIAL EQUATIONS, PHENIX-HALL, ENGLEWOOD CLIFFS,
 C 34 N.J., 1971.
 C 35
 C 36 PACKAGE ROUTINES CALLED... NONE
 C 37 USLW ROUTINES CALLED... NONE
 C 38 CALLED BY... STIFFIB
 C 39 FURTHER FUNCTIONS USED... FLOAT
 C 40
 C 41 DIMENSION PERST(112*2+3),EL(112),T(112)
 C 42 DATA PERST / 1.0,1.0,2.0,1.0, 315.0, .07407, .01391, .002182,
 C 43 * .0002945, .00014492, .000003642, .0000003524,
 C 44 * 1.0,1.0,5.1667, .01167, 1.0,1.0,1.0,1.0,1.0,
 C 45 * 2.0,2.0,2.0,37.89, 3.33,70.08,87.97,106.9,
 C 46 * 126.7,147.4,168.8,191.0,
 C 47 * 2.0,4.5,7.333,10.4213,1.0,1.0,1.0,1.0,1.0,
 C 48 * 12.0,24.0,37.89, 3.33,70.08,67.97,106.9,
 C 49 * 126.7,147.4,168.8,191.0,1,
 C 50 * 3.0,0.0,9.167,12.5,1.0,1.0,1.0,1.0,1.0,1.0, /
 C 51
 C 52 GO TO (1*?) *METH
 C 53 1 GO TO (110,102,103,104,105,106,117,108,109,110,111,112)*110
 C 54 2 GO TO (201,202,203,204,205)*NU
 C 55
 C 56 THE FOLLOWING COEFFICIENTS SHOULD BE DEFINED TO MACHINE ACCURACY.
 C 57 FOR A GIVEN ORDER NO, THEY CAN BE CALCULATED BY USE OF THE
 C 58 GENERATING POLYNOMIAL L(1), WHOSE COEFFICIENTS ARE EL(1) ...
 C 59 L(1) = F(1) + F(2)* ... + F(NU+1)*10^(NU).
 C 60 FOR THE IMPLICIT ADAMS METHODS, L(1) IS GIVEN BY
 C 61 DLDT = (1+1)*(1+2)* ... *(1+NU-1)/K, L(-1) = 0.
 C 62 WITH K = FACTORIAL((NU-1)).
 C 63 FOR THE RKF METHODS,

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SUBROUTINE COSET          76/76   OPT=1 KOUNT=--*/ 1-ACT      FIN 4.0K+49H
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C L(1) = (T+1)*(1+2)*...*(T+NU)/K.          59
C WHERE K = FACTORIAL(NU)*(1 + 1/2 + ... + 1/NU).
C
C THE ORDER IN WHICH THE GROUPS APPEAR BELOW IS..
C IMPLICIT ADAMS METHODS OF ORDERS 1 TO 12.
C RHF METHODS OF ORDERS 1 TO 5.
C
C
101 EL(1) = 1.0E-00
       GO TO 900
102 EL(1) = 0.5E-00
       EL(3) = 0.5E-00
       GO TO 900
103 EL(1) = 4.16666666666667E-01
       EL(3) = 0.75E-00
       EL(4) = 1.65666666666667E-01
       GO TO 900
104 EL(1) = 0.375E-00
       EL(3) = 9.16666666666667E-01
       EL(4) = 3.33333333333333E-01
       EL(5) = 4.16666666666667E-02
       GO TO 900
105 EL(1) = 3.48611111111111E-01
       EL(3) = 1.04166666666667E-00
       EL(4) = 4.86111111111111E-01
       EL(5) = 1.04166666666667E-02
       EL(6) = 8.33333333333333E-03
       GO TO 900
106 EL(1) = 3.29861111111111E-01
       EL(3) = 1.14166666666667E-00
       EL(4) = 0.625E-00
       EL(5) = 1.77083333333333E-01
       EL(6) = 0.025E-00
       EL(7) = 1.38888888888889E-03
       GO TO 900
107 EL(1) = 3.1559193121693E-01
       EL(3) = 1.725E-00
       EL(4) = 7.51951951951951E-01
       EL(5) = 2.55206333333333E-01
       EL(6) = 4.86111111111111E-02
       EL(7) = 4.86111111111111E-03
       EL(8) = 1.984126941270E-04
       GO TO 900
108 EL(1) = 3.0422453703704E-01
       EL(3) = 1.29642457142486E-00
       EL(4) = 6.6851651651652E-01
       EL(5) = 3.35763944444444E-01
       EL(6) = 7.77777777777778E-02
       EL(7) = 1.0648144144144E-02
       EL(8) = 7.93650793650793E-04
       EL(9) = 2.4401547301547E-05
       GO TO 900
109 EL(1) = 2.944640044094E-01
       EL(3) = 1.35942457142486E-00
       EL(4) = 9.7655423242423E-01
       EL(5) = 0.4171675E-00
       EL(6) = 1.11356165616561E-01
       EL(7) = 0.01675E-00

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FIN 4.8.0498

SUBROUTINE	COST	T	76/76	0PT=1 MOUND=--*/ TRACE	04/15/80	11-10.37	PAGE	53
115			$EL(8) = 1.9345238095638E-03$			116		
			$EL(9) = 1.11607142857145E-04$			117		
			$EL(10) = 2.7557319223986E-06$			118		
		GO TO 900	$EL(11) = 2.8697544642657E-01$			COSE	119	
120	110		$EL(12) = 1.414484126984E-00$			COSE	120	
			$EL(13) = 1.077215604656E-00$			COSE	121	
			$EL(14) = 4.9856701940035E-01$			COSE	122	
			$EL(15) = 0.1484375E-00$			COSE	123	
			$EL(16) = 2.9060570987054E-02$			COSE	124	
125			$EL(17) = 3.7202380952981E-03$			COSE	125	
			$EL(18) = 2.99685446550085E-04$			COSE	126	
			$EL(19) = 1.3778659611993E-05$			COSE	127	
			$EL(20) = 2.7557319223986E-07$			COSE	128	
		GO TO 900	$EL(21) = 2.8018959644994E-01$			COSE	129	
130	111		$EL(22) = 1.464484126984E-00$			COSE	130	
			$EL(23) = 1.1715145502646E-00$			COSE	131	
			$EL(24) = 5.7935819003527E-01$			COSE	132	
			$EL(25) = 1.8832286155203E-01$			COSE	133	
			$EL(26) = 4.1630362654421E-02$			COSE	134	
135			$EL(27) = 6.2111441798942E-03$			COSE	135	
			$EL(28) = 6.2520667989419E-04$			COSE	136	
			$EL(29) = 4.0417401528613E-05$			COSE	137	
			$EL(30) = 1.5156525573192E-06$			COSE	138	
			$EL(31) = 2.5052108383442E-08$			COSE	139	
		GO TO 900	$EL(32) = 2.7426554003160E-01$			COSE	140	
140	112		$EL(33) = 1.5099386124387E-00$			COSE	141	
			$EL(34) = 1.2602711640212E-00$			COSE	142	
			$EL(35) = 6.59234182094877E-01$			COSE	143	
145			$EL(36) = 2.3065800264550E-01$			COSE	144	
			$EL(37) = 5.5697246105232E-02$			COSE	145	
			$EL(38) = 9.4394841269441E-03$			COSE	146	
			$EL(39) = 1.1192749669312E-03$			COSE	147	
150			$EL(40) = 9.0939153434153E-05$			COSE	148	
			$EL(41) = 4.8225304641975E-06$			COSE	149	
			$EL(42) = 1.5031265031465E-07$			COSE	150	
			$EL(43) = 2.0876756987868E-09$			COSE	151	
155	201		GO TO 900	$EL(44) = 1.0E-00$		COSE	152	
			$EL(45) = 6.666666666666667E-01$			COSE	153	
			$EL(46) = 3.33333333333333E-01$			COSE	154	
			$EL(47) = 5.454545454545455E-01$			COSE	155	
160	202		$EL(48) = EL(1)$			COSE	156	
			$EL(49) = 9.09090404040491E-02$			COSE	157	
			$EL(50) = 0.02E-00$			COSE	158	
		GO TO 900	$EL(51) = 4.3795620437956E-01$			COSE	159	
165	204		$EL(52) = 0.48E-00$			COSE	160	
			$EL(53) = 0.7E-00$			COSE	161	
			$EL(54) = 0.2E-00$			COSE	162	
			$EL(55) = 0.02E-00$			COSE	163	
170	205		GO TO 900	$EL(56) = 8.2116704321168E-01$		COSE	164	
			$EL(57) = 3.10218497610219E-01$			COSE	165	
			$EL(58) = 1.10218497610219E-01$			COSE	166	
			$EL(59) = 3.10218497610219E-01$			COSE	167	
			$EL(60) = 8.2116704321168E-01$			COSE	168	
			$EL(61) = 3.10218497610219E-01$			COSE	169	
			$EL(62) = 1.10218497610219E-01$			COSE	170	
			$EL(63) = 3.10218497610219E-01$			COSE	171	
			$EL(64) = 8.2116704321168E-01$			COSE	172	

```

SUBROUTINE COSET    76/76   OPT=1  WOULD=---*/ 1-WACT
                           FIN 4. H444H   04/15/80  11.10.37
                           DATE

C      TL(5) = 5.4744525547445E-12
          TL(6) = 3.6496350364464E-03
          C      90U  WO 910 K = 1.3
          91U  10IK) = PENTSTIND*METHOK)
          TU(4) = .5E-004(2)/ FLAUT(NW+r)
          RETURN
          END

```

CARD NR.	SEVERITY	DETAILS	DIAGNOSIS OF PROBLEM
46	I	AN IF STATEMENT MAY HE MORE EFFICIENT THAN A FOR WHICH COMPUTED GO TO STATEMENT.	

SUBROUTINE DECS 76/76 OPT=1 MOUND=***/ TRACE FIN 4.0.H-44H 04/15/80 11.10.37 PAGE 56
 MX = 1 XM = ABS(B(I+1))
 CONTINUE
 40 IPIV(MX) = MX
 IF (MX .EQ. NM) GO TO 69
 DO 50 I = 1,LL
 XX = R(NK+I)
 R(NR+I) = R(IMX+I)
 R(IM+I) = XX
 50 XM = H(NR+I)
 60 IF (XM .EQ. 0.) GO TO 100
 H(NK+I) = 1./XM
 IF (ML .EQ. 0) GO TO 90
 XM = -B(NK+I)
 NK = MIND(N-NR,LL-1)
 DO 60 I = NP+LK
 J = LL + I - NR
 XX = R(I+1)*XM
 H(NK,J) = XX
 DO 70 II = 1,KK
 70 H(I+II) = H(I+II+1) + XX*(R(II+1))
 H0 H(I+LL) = 0.
 90 CONTINUE
 92 NR = N
 IF (H(N+1) .EQ. 0.) GO TO 100
 RETURN
 94 H(N+1) = 1./R(N+1)
 RETURN
 100 ICH = NM
 RETURN
 EN1

```

1      SUBROUTINE SOLB (NIM, N, ML, MU, R, Y, IPIV)
2      C SUBROUTINES DECB AND SOLB FORM A TWO SUBROUTINE PACKAGE FOR THE
3      C DIRECT SOLUTION OF A SYSTEM OF LINEAR EQUATIONS IN WHICH THE
4      C COEFFICIENT MATRIX IS REAL AND BANDL.
5
6      C SOLUTION OF A*X = C GIVEN LU DECOMPOSITION OF A FROM DECB.
7      C Y = RIGHT-HAND VECTOR C OF LENGTH N. ON INPUT.
8      C X = SOLUTION VECTOR X ON OUTPUT.
9      C ALL THE ARGUMENTS ARE INPUT ARGUMENTS.
10     C THE OUTPUT ARGUMENT IS Y.
11
12     C PACKAGE ROUTINES CALLED... NONE
13     C USER ROUTINES CALLED... DIFFUN,INITAL,STIF1B
14     C CALLED BY... MNG
15     C FURTHER FUNCTIONS USED...
16
17     C-----DIMENSION R(NDIM+1),Y(N),IPIV(N)
18
19     IF (N .EQ. 1) GO TO 60
20
21     N1 = N - 1
22     LL = ML + MU + 1
23     IF (ML .EQ. 0) GO TO 32
24     MU = 30 NR = 1,N1
25     IF (IPIV(NR) .EQ. NR) GO TO 10
26     J = IPIV(NR)
27     XX = Y(NP)
28     Y(NP) = Y(J)
29     Y(J) = XX
30     KK = MIN(N-NR+ML)
31     DO 20 I = 1,KK
32     Y(NR+I) = Y(NR+I) + Y(NR)*R(NR,LL+I)
33     CONTINUE
34     LL = LL - 1
35     Y(N) = Y(N)*R(N+1)
36     KK = 0
37     MU = 50 NB = 1,N1
38     NR = N - NH
39     IF (KK .NE. LL) KK = KK + 1
40     UP = 0.
41     IF (LL .EQ. 0) GO TO 50
42     DO 40 I = 1,KK
43     DP = DP + R(NR+I+1)*Y(NR+I)
44     Y(NR) = Y(NR) - UP*R(NR+1)
45     RETURN
46     Y(1) = Y(1)*R(1,1)
47     RETURN
48

```

*** 03/10/80 SCORT 2-1-5 H & L VER 00H *** 04/15/-11 H0106
 SYS IN VICTS H14V 4/F FLS=377K FLL=1750K MXS=300K MXL=131K MXH=1305H
 MH-NP-55 CPU SECOND OWNER
 11.09.36.MFA. A2 HHL MISHE 1.3 L499 VEN 004 03/17/80 PLFLSC.CY=1.
 12.47.45 00000.005 MFZ. -THWY-STWFZ.1200,P1.
 12.47.45 00000.005 JOR. -ACCOUNT.P100.
 12.47.45 00000.028 JORH. -MAP(UFF)
 12.47.46 00000.028 JORH. -ATTACH TAPE11.1WH20A1H3U11=SHLJMH.
 12.47.46 00000.032 MFZ. PF254 - CYCLE 2 ATTACH 1 FROM SNSYSTEM
 12.47.46 00000.032 JORH. -ATTACH.RFSC.HHZUCHTW30N2L11=SHLJMH.
 12.47.46 00000.036 MFZ. PF254 - CYCLE 1 ATTACH 1 FROM SNSYSTEM
 12.47.46 00000.037 JORH. -REQUEST TAPE2.0PT.
 12.47.46 00000.039 JORH. -REQUEST TAPE3.0PT.
 12.47.46 00000.040 JORH. -ATTACH.COLD.FLSL10=SHLJMH.
 12.47.46 00000.044 JORH. PF254 - CYCLE 5 ATTACH 11 FROM SNSYSTEM
 12.47.46 00000.045 JORH. -LOAD(COLD,MAIN)
 12.47.46 00000.048 JORH. -LOAD(COLD,INIT)
 12.47.46 00000.049 JORH. -LOAD(COLD,INIT)
 12.47.46 00000.050 JORH. -LOAD(COLD,FLSP)
 12.47.46 00000.051 JORH. -LOAD(COLD,HANDY)
 12.47.46 00000.051 JORH. -LOAD(COLD,BKPT)
 12.47.46 00000.051 JORH. -LOAD(COLD,PERCUL)
 12.47.46 00000.052 JORH. -LOAD(COLD,STIFIB)
 12.47.46 00000.052 JORH. -LOAD(COLD,INITIAL)
 12.47.46 00000.053 JORH. -LOAD(COLD,GFUN)
 12.47.46 00000.053 JORH. -LOAD(COLD,VALUES)
 12.47.46 00000.054 JORH. -LOAD(COLD,COLPN1)
 12.47.46 00000.054 JORH. -LOAD(COLD,HSPLYN)
 12.47.46 00000.055 JORH. -LOAD(COLD,INTERV)
 12.47.46 00000.055 JORH. -LOAD(COLD,EVAL)
 12.47.46 00000.056 JORH. -LOAD(COLD,DIFFUN)
 12.47.46 00000.057 JORH. -LOAD(COLD,ADDA)
 12.47.46 00000.057 JORH. -LOAD(COLD,HESS)
 12.47.47 00000.058 JORH. -LOAD(COLD,PSERIN)
 12.47.47 00000.059 JORH. -LOAD(COLD,DIFFET)
 12.47.47 00000.059 JORH. -LOAD(COLD,INTEMP)
 12.47.47 00000.060 JORH. -LOAD(COLD,CUST1)
 12.47.47 00000.061 JORH. -LOAD(COLD,DECH)
 12.47.47 00000.061 JORH. -LOAD(COLD,SULH)
 12.47.47 00000.061 JORH. -EXECUTE.
 12.47.53 00000.440 MFZ. LD610 - FLS REQUIRED TO LOAD - QUIT1323 00.C06
 12.47.54 00000.442 MFZ. LD603 - EXECUTION INITIATED 05.LAP
 12.47.54 00000.442 USH. FURTHER LIHART 498 01/03/M0
 12.50.20 00034.949 USR. STOP
 12.50.20 00034.949 USP. 34.006 CP SECONDS EXECUTION TIME.
 12.50.21 00034.949 JORH. -CATALOG TAPE2.H20A1H3U11+1D=SHLJMH.
 12.50.21 00034.950 MFZ. PF060 - CYCLE 3 CATALOG+D UN SNSYSTEM
 12.50.21 00034.950 JORH. -CATALOG TAPE3.H20A1H3U11+1D=SHLJMH.
 12.50.21 00034.953 MFZ. PF060 - CYCLE 3 CATALOG+D ON SNSYSTEM
 12.50.21 00034.957 MFZ. JM166 - MAXIMUM USEN SCM 111111 WORDS
 12.50.21 00034.961 MFZ. JM167 - MAXIMUM USEN LCM 00 WORDS
 12.50.21 00034.961 MFZ. JM170 - MAXIMUM JS+IU LCM 173 WORDS
 12.50.21 00034.961 MFZ. JM771 - OPEN/CLOSE CALLS
 12.50.21 00034.961 MFZ. JM772 - DATA TRANSFM CALLS
 12.50.21 00034.961 MFZ. JM773 - CINTROL/POSITION1... CALLS
 12.50.21 00034.961 MFZ. JM774 - HM LUTIN TRANSFM CALLS
 12.50.21 00034.961 MFZ. JM775 - HM LUTIN POSITION1 IN CALLS
 12.50.21 00034.961 MFZ. JM776 - HM LUTIN POSITION2 CALLS

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H2-AIR FLAME. 50 PERCENT H2. WARMATZ KINETICS.

12.50.21	00034.468	MF2.	HM777 - RECALL CALLS	1	317.071	KWS
12.50.21	00034.469	MF7.	SCM	317.071	KWS	
12.50.21	00034.869	MF2.	LCM	317.071	KWS	
12.50.21	00034.869	MF2.	I/O	0.017	MWS	
12.50.21	00034.469	MF2.	RMS	0.028	MWS	
12.50.21	00034.469	MF2.	USER	34.297	St.C	
12.50.21	00034.470	MF2.	JOH	34.017	St.C	
12.50.21	00034.470	MF2.	D10	311.314	K4	
12.50.21	00034.470	MF2.	SS	45.174	SEC	
12.50.21	00034.470	MF2.	COST ESTIMATE	56.949		
12.50.21	00034.470	MF2.	SC050 - 000015	SC/LC Swap		

PMESSUMt = 1.00000E+00 ATM.

NPUDE = 4

		W	UC	UH	HU	H2D
1	H	1.0000E+00	6.44851E-14	5.2691E-02	5.2100E+04	4.5434E-07
2	UH	1.1025E+01	5.2566E+01	5.5839E+02	1.3530E+07	
3	0	1.6000E+01	1.0376E+10	1.0469E+07	3.7216E+03	1.3746E+07
4	H02	3.3900E+01	2.1401E+10	1.9533E+11	1.5166E+02	9.0984E+08
5	H2O2	3.4000E+01	2.2044E+10	1.3187E+10	-9.5663E+02	9.0471E+08
6	H2	2.0000E+00	6.44851E-02	3.7564E+02	1.5951E+01	2.3005E+07
7	U2	3.2000E+01	2.1790E+01	4.3495E+04	-3.5695E+02	9.6265E+08
8	H2O	1.8000E+01	1.673E+00	2.4508E+01	-3.210E+03	1.1638E+07
9	N2	2.8000E+01	7.1725E+01	7.1725E+01	-3.9514E+02	1.3902E+07

TC = 2.9200E+01 TH = 1.9366E+00

TPN = 1.00000E+03 PHN = 5.00000E-05 TMN = 1.00000E-03

CPNA = 4.7850E-01 HL = 7.0440E-08 TH..1 = 2.9490E+02

PHD = 0. PHB = 6.00000E-01 TH..1..L = 5.00000E-01

EPS = 1.00000E-03 SHC = 1.00000E-06

NSTANT = 1 NINT = 12 NMJMH = 0 PLUSH = 0.

NTHAI = 1 THINT = 5.00000E-01

THAA = 1.00000E+02

NCHI = 4 HC = 4.00000E+01

INITIAL DENSITY = 0.3065F-04

0.	9.8011E-02	1.6733E-01	2.1634E-01	2.5044E-01	2.7550E-01	3.0000E-01	3.2450E-01	3.4901E-01	3.4366E-01
4.3247E-01	5.0198E-01	6.0000E-01	7.4505E-02	3.6757E-02	4.9009E-02	6.1261E-02	7.3514E-02	9.5166E-02	9.4014E-02
1.2252E-02	1.2401E-01	1.3267E-01	1.4139E-01	1.5000E-01	1.5866E-01	1.6733E-01	1.7545E-01	1.7958E-01	1.8571E-01
1.1535E-01									
1.9143E-01									
2.4233E-01									
2.7550E-01									
3.0613E-01									
3.3678E-01									
3.7560E-01									
4.3267E-01									
5.2649E-01									

KCEN = 9 VCEN = 9.5343E-01

T = 0. DT = 0. TOTAL STEPS = 0

NF = 0 NJ = 0

MUN TIME = 0.0000E-03

M2D AT LEFT = 9.0M67E-03 2.7060E-03 2.7442E-03 1.6114E-03 1.6994E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

UPH AT LEFT = 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.

UPH? AT LEFT = 2.0680E-25 0. 0. 0. 0. 0. 0. 0. 0. 0.

FLAME SPOTU = -2.5516E+03 -R.6066E+02 2.2673E+02 -2.6111E+10 -3.7377E+06 1.6050E+012 R.H274E+01 R.2698E+01 6.4080E+09

FLAME SPOTU = 7.6276E+02 -2.4726E+04 3.1661E+06 -6.7844E+09 -7.2197E+05 5.3024E+01 2.7720E+01 3.0761E+01 1.8349E+09

FLAME SPOTU = R.3493E+02 -2.5153E+04 4.6471E+06 -5.7314E+09 -7.1461E+05 R.0253E+00 4.6144E+00 8.0431E+00 0.

FLAME FRONT FRM PH1# = 2.5095E+01 TU PH1# = 1.5759E+01

FLAME FRONT FRM A = 2.0946E-02 TU X = 4.0460.11-02

FLAME THICKNESS = 2.7577E-02 CM

NU = 5.0724E-03 FLSP = R.0431E+00

SPEEHD = -1.0145E-03 DS = 0.

T = 5.0000E-03 DI = 2.7511E-04 TOTAL STEPS = 51

HUN TIME = 5.8600t+00

W2D AT LEFT = 9.0467t-03 2.7060t-03 2.7042t-03 1.8147t-03 1.8094t-03 4.6010t-03 1.8849t-03 2.3677t-03 0.

UPH AT LEFT = 1.3411t-09 -1.7184t-09 5.4406t-10 5.1574t-09 6.2104t-09 -4.3164t-06 -3.1783t-05 1.4965t-05 1.4976t-05 7.6777t-05

UPH2 AT LEFT = -1.4652t-07 -5.3094t-07 -9.6154t-08 -3.8054t-05 -4.4577t-07 1.0648t-03 -6.4661t-03 -1.8076t-03 -3.8844t-02

FLAME SPEEDU = 1.7634t+02 1.9839t+02 4.5745t+02 2.8354t+02 1.3178t+02 2.0476t+02 4.6679t+01 1.1624t+02 6.6176t+02

FLAME SPEEDU = 5.9500t+02 4.6934t+02 7.5362t+02 6.6571t+02 5.3483t+04 7.3938t+01 3.8262t+01 4.3247t+01 -H.1220t+02

FLAME SPEEDU = 2.6660t+03 1.9866t+03 6.5519t+05 -4.1572t+09 -1.3298t+08 2.0165t+01 1.6072t+01 1.4593t+01 -3.0946t+02

FLAME FRONT FROM PHI0 = 2.4225t-01 TO PHI0 = 1.6627t-01

FLAME FRONT FROM X = 2.0267t-02 TO X = 5.2612t-02

FLAME THICKNESS = 3.2565t-02 CM

M0 = 3.8372t-02 FLSP = 6.0845t+01 PHNEW = 2.9967t-01

SPN1 = -7.6744t-01 SPN2 = -1.5648t+00 SPEED1 = -7.6744t-01 US = -6.6610t-01

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T = 1.0000E-02 DT = 6.3976t-04 TOTAL STEPS = 62

NF = 64 NJ = 9

HUN TIME = 6.4270t+00

W2D AT LEFT = 9.0467t-03 2.7060t-03 2.7042t-03 1.8147t-03 1.8094t-03 4.6010t-03 1.8849t-03 2.3677t-03 0.

UPH AT LEFT = 4.0121t-09 8.8467t-09 5.6945t-09 1.3841t-07 3.1864t-07 -2.6803t-05 -3.7487t-05 6.0286t-05 3.8157t-06 H.3990t-04

UPH2 AT LEFT = -5.5150t-07 -1.8045t-06 -4.8160t-07 -6.2574t-05 -4.5260t-06 4.9272t-03 4.4479t-03 -1.2060E-02 -1.2443t-03 -1.4961t-01

FLAME SPEEDU = 2.5450t+02 3.5339t+02 4.0442t+02 3.0417t+02 3.0110t+02 1.7720t+012 4.3H57t+01 1.0844E+02 3.5161t+02

FLAME SPEEDU = 5.4617t+02 4.8131t+02 6.7111t+02 5.4544t+03 4.3531t+04 4.6939t+01 5.1043E+01 5.5297t+01 4.1936t+02

FLAME SPEEDU = 5.4566t+03 4.2421t+03 1.2790t+06 5.9311t+00 4.6255t+00 4.5591E+01 5.5784t+01 3.1671E+01 2.4146t+02

FLAME FRONT FROM X = 2.00047t-02 T, X = 5.042t-012

FLAME THICKNESS = 3.0374E-02 CM

W1 = 2.0700E-02 U1 LSP = 4.5509E+01 PHASEW = 2.4693E-01

SPN1 = -5.7600E-01 SPN2 = -6.2000E-01 SPt1,11 = -5.7600E-01 DS = 1.0304E-01

T = 1.0540E-02 DT = 7.6771E-04 TOTAL STRSS = 11

NF = 111 NJ = 11

RUN TIME = 9.0630E+00

W2D AT LEFT = 9.0467E-03 2.7060E-03 < 7.642E-03 1.0147E-03 1.8094E-03 4.0101E-03 1.0449E-03 2.1677E-03 0.

UPH AT LEFT = 1.0160E-04 6.1181E-08 4.5051E-08 -5.4591E-08 -7.1372E-05 -1.0440E-04 1.7400E-04 -2.4770E-06 2.1453E-03

UPH2 AT LEFT = -2.5080E-06 -1.0467E-05 -7.0314E-06 -6.0100E-05 -2.0656E-05 1.1245E-02 1.0853E-02 -3.2268E-02 2.2768E-03 -4.0004E-01

FLAME SPt1,0 = 2.3171E+02 3.1362E+02 3.6517E+02 3.2171E+02 4.7272E+02 1.5612E+02 1.0011E+02 1.0781E+02 2.2795E+02

FLAME SPt1,0 = 4.4120E+02 4.5064E+02 5.6711E+02 5.0414E+03 2.4456E+05 1.2997E+02 9.1541E+01 9.4450E+01 -2.4342E+02

FLAME SPt1,0 = 1.4396E+04 8.8076E+03 1.9604E+00 3.2011E+08 -6.0778E+08 1.2016E+02 -4.6871E+01 9.0H39E+01 1.5316E+02

FLAME + FRONT FRUM PH1,0 = 2.2925E-01 TU PH1,0 = 1.9590E-01

FLAME FRONT FRUM X = 1.9410E-02 TU X = 6.5311E-02

FLAME THICKNESS = 4.5901E-02 CM

W1 = 3.7339E-02 U1 LSP = 5.9206E+01 PHASEW = 2.9433E-01

SPN1 = -7.4677E-01 SPN2 = -1.1800E+00 SPt1,11 = -7.4677E-01 DS = -1.7277E-01

T = 3.1390E-02 DT = 8.0203E-04 TOTAL STRSS = 45

NF = 143 NJ = 13

RUN TIME = 1.1249E+02

W2D AT LEFT = 9.0467E-03 2.7060E-03 < 7.642E-03 1.0147E-03 1.8094E-03 4.0101E-03 1.0449E-03 2.1677E-03 0.

UPH AT LEFT = 1.3616E-04 4.0264E-07 7.0554E-07 -1.0251E-05 2.0034E-06 -5.0112E-05 2.0034E-06 4.0009E-04 1.0774E-04 9.0526E-04

$UPH2 \text{ AT LEFT} = -1.0791t^{-0.15} - 7.3639t^{-0.05} - 9.0405t^{-0.05} 8.07mAE^{-0.04} - 3.7392t^{-0.04} 1.05602t^{-0.02} 4.0314t^{-0.02} - 6.7402t^{-0.02} 1.1674t^{-0.02} - 7.3979t^{-0.01}$
 $FLAME SPTEU = 2.02109t^{-0.02} 3.3001t^{-0.02} 3.0465t^{-0.02} 1.4563t^{-0.02} 5.0884t^{-0.02} 1.04642t^{-0.02} 1.0393t^{-0.02} 1.03715t^{-0.02} 1.0403t^{-0.02}$
 $FLAME SPTEU = 3.0230t^{-0.02} 3.7075t^{-0.02} 2.0901t^{-0.02} 6.2742t^{-0.02} 1.07424t^{-0.02} 1.0743t^{-0.02} 1.05547t^{-0.02} 1.05900t^{-0.02} - 0.35005t^{-0.01}$
 $FLAME SPTEU = 1.0330t^{-0.04} 5.02534t^{-0.03} 1.2735t^{-0.03} 1.0214t^{-0.02} 1.03214t^{-0.02} 1.0324t^{-0.02} 1.0055t^{-0.02} 1.0731t^{-0.02} 3.05007t^{-0.01}$

$\text{FLAME FRONT FMU} \lambda = 1.04962t^{-0.02} \text{ Tu } x = 7.0531t^{-0.01} \text{ Tu } \mu H1 = 4.0162t^{-0.01}$
 $\text{FLAME FRONT FMU} \lambda = 2.1631t^{-0.01} \text{ Tu } x = 7.0531t^{-0.01}$

$\text{FLAME THICKNESS} = 5.04824t^{-0.02} \text{ cm}$

$MO = 7.7032t^{-0.02} \text{ FLSP} = 1.02215t^{-0.02} \text{ muH1} = 2.0102t^{-0.01}$

$SPN1 = -1.5407t^{-0.00} \text{ SPN2} = -5.05740t^{-0.00} \text{ SPN3} = -3.05656t^{-0.00} \text{ DS} = -2.01131t^{-0.00}$

$T = 3.0342t^{-0.02} \text{ DT} = 8.0527t^{-0.04} \text{ TOTAL } L12t^{-0.05} = 109$

$NF = 169 \text{ NJ} = 15$

$\text{RUN TIME} = 1.03591t^{-0.01}$

$W2D \text{ AT LEFT} = 9.0467t^{-0.03} 2.7060t^{-0.03} 2.7442t^{-0.03} 1.0117t^{-0.03} 1.00066t^{-0.03} 4.00010t^{-0.03} 1.00009t^{-0.03} 2.03677t^{-0.03} 0.$
 $UPH \text{ AT LEFT} = 2.05762t^{-0.07} 3.04657t^{-0.06} 3.04319t^{-0.06} - 0.00064t^{-0.05} 1.00302t^{-0.05} - 0.0005792t^{-0.04} 9.03984t^{-0.04} 1.05412t^{-0.04} 1.0331t^{-0.02}$
 $UPH2 \text{ AT LEFT} = -3.04347t^{-0.05} - 3.08602t^{-0.04} - 0.00068t^{-0.04} 0.01111t^{-0.03} - 1.02046t^{-0.03} 1.05402t^{-0.02} 1.09617t^{-0.02} - 1.1364t^{-0.01} 1.04161t^{-0.01} - 1.00008t^{-0.00}$
 $FLAME SPTEU = 2.02151t^{-0.02} 2.04369t^{-0.02} 2.06530t^{-0.02} 1.00414t^{-0.02} 4.02863t^{-0.02} 1.7817t^{-0.02} 1.05631t^{-0.02} 1.05835t^{-0.02} 1.04004t^{-0.02}$
 $FLAME SPTEU = 2.04661t^{-0.02} 3.03914t^{-0.02} 4.16294t^{-0.02} - 1.0219t^{-0.01} 4.000375t^{-0.01} 2.06448t^{-0.02} 1.08109t^{-0.02} 1.08457t^{-0.02} - 0.014147t^{-0.01}$
 $FLAME SPTEU = 7.0190t^{-0.03} 3.01005t^{-0.03} 2.05712t^{-0.05} - 0.0006611t^{-0.07} 1.0219t^{-0.06} 2.01504t^{-0.02} 1.08278t^{-0.02} 1.04023t^{-0.02} 0.04744t^{-0.00}$
 $\text{FLAME FRONT FMU} \lambda = 2.1631t^{-0.01} \text{ Tu } \mu H1 = 4.01325t^{-0.01}$
 $\text{FLAME FRONT FMU} \lambda = 1.04943t^{-0.02} \text{ Tu } x = 4.0140t^{-0.02}$
 $\text{FLAME THICKNESS} = 6.03364t^{-0.02} \text{ cm}$

$MO = 1.01640t^{-0.01} \text{ FLSP} = 1.08541t^{-0.01} \text{ muH1} = 2.04562t^{-0.01}$
 $SPN1 = -2.03291t^{-0.00} \text{ SPN2} = -0.04543t^{-0.00} \text{ DS} = -0.03162t^{-0.00} \text{ NF} = -0.01611t^{-0.00} \text{ NF1} = 0.0000t^{-0.00}$

$t = 4.6021E-02$ $dt = 6.3344E-04$ $\text{TOTAL STEPS} = 123$
 $NF = 190$ $NJ = 17$
 $\text{RUN TIME} = 1.5322E+01$

$W2D \text{ AT LEFT} = 9.0667E-03$ $2.7060E-03$ $2.7492E-03$ $1.617E-03$ $1.8094E-03$ $4.6010E-03$ $1.4449E-03$ $2.7677E-03$ $0.$
 $UPH \text{ AT LEFT} = 1.6379E-07$ $5.9733E-06$ $1.0218E-06$ $-1.4571E-04$ $1.0728E-05$ $-2.9R31E-04$ $-1.0717E-03$ $1.2402E-03$ $2.5444E-04$ $1.0339E-02$
 $UPH2 \text{ AT LEFT} = -2.1751E-05$ $-6.9877E-04$ $-4.4307E-04$ $1.4547E-02$ $-1.0355E-02$ $1.3777E-02$ $1.3944E-01$ $-1.3197E-01$ $-3.0135E-03$ $-1.0641E+00$
 $FLAME SPEEDU = 2.2124E+02$ $2.0829E+02$ $2.4144E+02$ $1.4614E+02$ $3.1293E+02$ $1.8574E+02$ $1.6963E+02$ $1.0967E+02$ $1.7904E+02$
 $FLAME SPEEDU = 2.7012E+02$ $3.1793E+02$ $3.3649E+02$ $-1.0571E+03$ $2.8065E+05$ $2.1727E+02$ $1.9K36E+02$ $2.0087E+02$ $-1.9702E+02$
 $FLAME SPEDU = 3.7140E+03$ $1.5632E+03$ $6.7016E+04$ $-6.4575E+06$ $2.4318E+07$ $2.3393E+02$ $<0.007E+02$ $1.9708E+02$ $3.7685E+00$
 $FLAME FRONT FWHM PHI* = 2.2497E-01$ $10 \text{ PHI*} = 4.5854E-01$
 $FLAME FRONT FWHM X = 1.9091E-02$ $TU X = 4.08014E-02$
 $FLAME THICKNESS = 6.9724E-02$ CW
 $W0 = 1.3188E-01$ $FLSP = 2.0912E+02$ $PHNEW = 2.9280E-01$
 $SPN1 = -2.6376E+00$ $SPN2 = -2.4335E+00$ $SPTL1,1 = -2.7350E+00$ $DS = 1.1429E+00$
 $t = 5.6025E-02$ $dt = 1.0479E-03$ $\text{TOTAL STEPS} = 134$
 $NF = 209$ $NJ = 20$
 $\text{RUN TIME} = 1.7066E+01$

$K2D \text{ AT LEFT} = 9.0667E-03$ $2.7060E-03$ $2.7492E-03$ $1.617E-03$ $1.8094E-03$ $4.6010E-03$ $1.4449E-03$ $2.3677E-03$ $0.$
 $UPH \text{ AT LEFT} = 9.1674E-04$ $2.9204E-06$ $4.5704E-06$ $-1.3637E-04$ $1.4224E-05$ $-2.2250E-04$ $-1.9684E-04$ $7.5415E-04$ $3.7922E-04$ $6.0855E-03$
 $UPH2 \text{ AT LEFT} = -1.2341E-05$ $-3.6947E-04$ $-5.7417E-04$ $1.2217E-02$ $-1.5432E-03$ $9.1537E-03$ $1.9134E-02$ $-4.2616E-02$ $-1.6102E-02$ $-6.5492E-01$
 $FLAME SPEEDU = 2.3534E+012$ $2.3211E+002$ $2.3325E+012$ $1.9400E+012$ $1.7013E+012$ $1.4744E+012$ $1.9006E+012$ $1.8114E+012$
 $FLAME SPEDU = 2.7566E+012$ $2.9605E+012$ $3.0435E+012$ $-4.0314E+012$ $1.6643E+015$ $2.3271E+012$ $2.0206RF+012$ $4.7164E+012$ $2.2164E+012$

FLAME FRONT FROM PH1 = 2.2916E-01 TO PH1* = 4.4324E-01

FLAME FRONT FROM X = 1.9226E-02 TO X = 4.4404E-02

FLAME THICKNESS = 8.0153E-02 CM

NU = 1.4305E-01 FLS = 2.2683E-02

PHNEM = 2.9724E-01

SPN1 = -2.04610E+00 SPN2 = -3.3409E+00

SPTEU = -2.04610E+00

T = 7.1604E-02 U1 = 1.4736E-03 TOTAL SPTEU = 154

NF = 23 NJ = 25

XUN TIME = 1.9431E+01

W20 AT LEFT = 9.0667E-03 2.7060E-03 2.7492E-03 1.6191E-03 1.6094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

WPH AT LEFT = 1.2103E-07 2.0517E-06 3.6109E-06 -1.0434E-04 2.2467E-05 -1.9511E-04 -3.6975E-04 3.7517E-04 2.6495E-04 4.2778E-03

WPH AT LEFT = -1.6295E-05 -3.7116E-04 -4.8423E-04 1.0496E-02 -2.0414E-03 6.8793E-03 3.9632E-02 -4.5179E-02 -4.5144E-03 -4.6556E-01

WPH AT LEFT = -1.6295E-05 -3.7116E-04 -4.8423E-04 1.0496E-02 -2.0414E-03 6.8793E-03 3.9632E-02 -4.5179E-02 -4.5144E-03 -4.6556E-01

FLAME SPTEU = 2.5372E+02 2.6206E+02 2.4767E+02 1.1310E+02 2.7471E+02 2.1963E+02 2.2302E+02 2.1924E+02 2.0520E+02

FLAME SPTEU = 2.0464E+02 3.0762E+02 3.1457E+02 -1.1501E+03 1.0114E+03 2.5447E+02 2.4909E+02 2.4486E+02 2.4490E+02

FLAME SPTEU = 9.3604E+02 5.4921E+02 3.1791E+03 -7.0254E+05 8.1194E+05 2.9265E+02 2.5244E+02 2.5281E+02 -5.0441E+00

FLAME FRONT FROM PH1 = 2.2925E-01 TO PH1* = 4.0176E-01

FLAME FRONT FROM X = 1.9137E-02 TO X = 4.0254E-02

FLAME THICKNESS = 8.3358E-02 CM

NU = 1.5444E-01 FLS = 2.4575E+02

PHNEM = 2.9254E-01

SPN1 = -3.0997E+00 SPN2 = -4.1240E+00

SPTEU = -3.0997E+00

DS = -1.5041E-01

T = 9.0143E-02 U1 = 1.4736E-03 TOTAL SPTEU = 170

NF = 23 NJ = 25

XUN TIME = 2.0414E+01

W2D AT LEFT = 9.0867E-03 P.7060F-03 2.7492E-03 1.8147E-03 1.8949E-03 4.6010E-03 1.0849F-03 2.3677E-03 0.
UPH AT LEFT = 1.3494E-07 5.7160F-06 5.9374E-06 -2.2213E-04 4.0230E-05 -1.3166E-04 -1.5909E-04 3.9561E-04 2.6359E-04 3.4990E-04

UPH AT LEFT = -1.0579E-05 -6.8406E-04 -7.1516E-04 2.1149E-02 -4.2580E-03 6.6517E-03 1.6656E-02 -4.3869E-02 -1.5466E-02 -4.0556E-01

*FLAME SPEED = 2.7093E+02 2.6654F+02 2.6395E+02 1.8491E+02 2.0821E+02 2.4401E+02 2.5285E+02 2.5002E+02 2.6116E+02

*FLAME SPEED = 2.8627E+02 2.8993E+02 3.0330E+02 -2.2949E+03 7.8673E+04 2.6895E+02 2.6902E+02 2.6790E+02 1.6602E+02

*FLAME SPEED = 5.6637E+02 4.3163E+02 1.2113E+03 -3.1134E+06 3.5088E+05 3.0557E+02 <7.086E+02 2.7380E+02 -6.0184E-01

FLAME FRONT FRM PH1* = 2.2924E+01 TU PH1* = 1.0191E-01

FLAME FRONT FRM X = 1.8953E-02 TU X = 1.0211E-01

FLAME THICKNESS = 8.3159E-02 CM

PH0 = 1.6643E-01 FLSP = 2.6390E+02 PHNEW = 2.9041E-01

SPN1 = -3.3286E+00 SPN2 = -3.9051E+00 SPt1,i,i = -3.5172E+00 NS = -5.2937E-03

T = 1.1127E-01 DT = 2.4007E-03 TOTAL S1,NS = 189

NF = 240 NJ = 24

HUN TIME = 2.3101F+01

W2D AT LEFT = 9.0867E-03 2.7060F-03 2.7492E-03 1.8147E-03 1.8949E-03 4.6010E-03 1.0849F-03 2.3677E-03 0.

UPH AT LEFT = 9.7204E-08 4.2032F-06 5.3071E-06 -2.0392E-04 5.9178E-05 -7.5909E-05 -1.5980E-04 3.0913E-04 2.5142E-04 2.7671E-03

UPH AT LEFT = -1.2921E-05 -6.9446E-04 -6.2235E-04 1.85259E-02 -5.6652E-03 5.4452E-03 5.5077E-02 -3.3260E-02 -1.8026E-02 -3.3763E-01

*FLAME SPEED = 2.9237E+02 2.2559E+02 2.6960E+02 2.5041E+02 2.9052E+02 2.6244E+02 2.6945E+02 2.7149E+02

*FLAME SPEED = 2.9687E+02 2.5857E+02 2.8790E+02 -2.5611E+02 2.4417E+04 2.7553E+02 2.7586E+02 2.1491E+02

*FLAME SPEED = 3.3624E+02 3.7477E+02 5.4240E+02 -3.9911E+03 2.5455E+03 2.9944E+02 2.7439E+02 2.4174E+02 3.4135E-01

FLAME FRONT FRM PH1* = 2.3360E+01 TU PH1* = 4.0202E-01

FLAME THICKNESS = 1.9196E-02 TU X = 6.0456E-01

FLAME THICKNESS = 4.5364E-02 CM

$\mu_0 = 1.7320E-01$ $FLSP = 2.7476E+012$ $P_{HNE} = 2.9405E-01$
 $SPN1 = -3.4656E+00$ $SPN2 = -3.8352E+00$ $SPET_{111..} = -3.46556E+00$ $D_S = 1.5154E-01$

$t = 1.3607E-01$ $DT = 2.4007E-03$ TOTAL $S_{111..S} = 149$
 $NF = 297$ $NJ = 24$

RUN TIME = $2.4247E+01$

$W2D AT LEFT = 9.0867E-03$ $2.7060E-03$ $2.7492E-03$ $1.8141E-03$ $1.8094E-03$ $4.6010E-03$ $1.0049E-03$ $2.3677E-03$ 0.
 $UPH AT LEFT = 4.7549E-04$ $3.1231E-06$ $4.1576E-06$ $-1.3014E-04$ $6.6511E-05$ $-5.2224E-05$ $-2.8264E-04$ $2.1760E-04$ $1.7154E-04$ $2.0367E-03$

$UPH^2 AT LEFT = -1.1677E-05$ $-3.7422E-04$ $-4.9485E-04$ $1.0113E-02$ $-6.5444E-03$ $4.6128E-03$ $2.7916E-02$ $-2.4007E-02$ $-1.3040E-02$ $-2.6151E-01$

FLAME SPETU = $3.0079E+02$ $2.1841E+02$ $2.7280E+02$ $2.6711E+02$ $2.9434E+02$ $2.7352E+02$ $2.7621E+02$ $2.7790E+02$ $2.4322E+02$
FLAME SPTEO = $2.6263E+02$ $2.5003E+02$ $2.6267E+02$ $5.0201E+01$ $5.0201E+04$ $2.7833E+02$ $2.7999E+02$ $2.7976E+02$ $2.6754E+02$
FLAME SPTEU = $3.2633E+02$ $3.1574E+02$ $3.0523E+02$ $-1.0911E+05$ $1.0687E+05$ $2.8950E+02$ $2.8019E+02$ $2.8173E+02$ $-5.4120E-01$

FLAME FRONT FNUM PH1# = $2.3354E-01$ TO PH1# = $3.8363E-01$

FLAME FRONT FNUM X = $1.4115E-02$ TO X = $5.7874E-02$

FLAME THICKNESS = $3.8754E-02$ C14

$MU = 1.7602E-01$ $FLSP^2 = 2.7912E+02$ $P_{HNEW} = 2.9453E-01$
 $SPN1 = -3.5205E+00$ $SPN2 = -3.0851E+00$ $SPET_{111..} = -3.7205E+00$ $D_S = -5.04910E-02$

RUN TIME = $2.5940E+01$

$W2D AT LEFT = 9.0867E-013$ $2.7059E-03$ $2.7492E-03$ $1.8141E-03$ $1.8094E-03$ $4.6010E-03$ $1.0049E-03$ $2.3677E-03$ 0.
 $UPH AT LEFT = 1.03117E-01$ $3.7514E-06$ $4.0433E-06$ $-1.5252E-05$ $-4.0393E-05$ $-1.0774E-04$ $1.0404E-04$ $2.27919E-04$ $1.47311E-04$ $2.0479E-01$

$UPH^2 AT LEFT = -1.3679E-05$ $-4.4793E-04$ $-7.3424E-04$ $1.4411E-02$ $-7.1571E-03$ $4.0614E-03$ $-2.7400E-02$ $-2.4793E-02$ $-1.3110E-02$ $-2.6520E-01$

FLAME SPEED = 2.972E+02 2.2471E+02 2.7451E+02 2.5977E+02 2.9639E+02 2.7920E+02 2.7795E+02 2.8046E+02 2.8795E+02
FLAME SPEED = 2.9156E+02 2.4841E+02 2.8220E+02 2.8220E+02 2.1520E+04 2.1520E+04 2.7981E+02 2.8036E+02 2.8046E+02 2.8336E+02
FLAME SPEED = 2.4209E+02 2.6262E+02 2.1471E+02 2.1471E+02 -1.9914E+05 1.4745E+05 2.6100E+02 2.6100E+02 2.8040E+02 2.8023E+02 2.7016E+00

FLAME FRONT FLUM PH1* = 2.3358E-01 TO PH1* = 3.9590E-01

FLAME FRONT FLUM X = 1.9145E-02 TO X = 6.2159E-02

FLAME THICKNESS = 4.3614E-02 CM

WT = 1.7690E-01 FLSP = 2.8050E+02 PHNT = 2.9295E-01

SPN1 = -3.5310E+00 SPN2 = -3.7245E+00 SPTEL1* = -3.6337E+00 DS = -1.1321E-01

T = 2.5039E-01 DT = 7.3361E-03 TOTAL STEPS = 231

NF = 346 NJ = 31

HUN TIME = 2.8076E+01

X2D AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8171E-03 1.6094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.
UPH AT LEFT = 7.7601E-04 2.9957E-06 4.2660E-06 -1.1401E-04 7.6444E-05 -3.7055E-05 -2.9985E-04 2.1739E-04 1.5474E-04 1.9192E-03
UPH2 AT LEFT = -1.0324E-05 -3.5239E-04 -5.0162E-04 1.01<0-02 -7.2342E-03 4.2275E-03 2.9310E-02 -2.3551E-02 -1.2709E-02 -2.4592E-01

FLAME SPEED = 3.0737E+02 2.1350E+02 2.7354E+02 2.6915E+02 2.9428E+02 2.4121E+02 2.7799E+02 2.8121E+02 2.8070E+02

FLAME SPEED = 2.6136E+02 2.4861E+02 2.6218E+02 -7.5124E+01 3.7126E+04 2.8059E+02 2.8037E+02 2.8060E+02 2.8733E+02

FLAME SPEED = 2.6339E+02 2.4157E+02 1.9220E+02 -1.1635E+05 1.0440E+05 2.8005E+02 2.8034E+02 2.8059E+02 -2.1515E+01

FLAME FRONT FLUM X = 1.9535E-02 TO X = 6.8854E-02

FLAME THICKNESS = 4.4118E-02 CM

WT = 1.7700E-01 FLSP = 2.8067E+02 PHNT = 2.9597E-01

SPN1 = -3.5411E+00 SPN2 = -3.5112E+01 SPTEL1* = -3.5401E+00 DS = 4.3644E-02
T = 3.7539E-01 DT = 5.6542E-03 TOTAL STEPS = 254

NF = 391 NJ = 36

RUN TIME = 3.1H10E+01

W20 AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8147E-03 1.8034E-03 4.6010E-03 1.0849E-03 2.3677E-03 0.

WPM AT LEFT = 4.3081E-08 1.5657E-06 2.6654E-06 -3.5304E-05 4.3792E-05 -2.0572E-05 -2.0169E-04 1.3041E-04 1.7095E-05 1.3275E-03

UPH2 AT LEFT = -5.738E-06 -1.8641E-04 -3.1743E-04 3.1411E-03 -4.1940E-03 3.2514E-03 1.9845E-02 -1.4329E-02 -7.2934E-03 -1.7179E-01

FLAME SPEED = 3.2213E+02 2.3411E+02 2.7500E+02 2.5611E+02 2.6298E+02 2.8117E+02 2.7743E+02 2.8234E+02 2.8054E+02

FLAME SPTEU = 2.8120E+02 2.6997E+02 2.8129E+02 -1.0304E+03 5.7945E+03 2.8065E+02 2.8043E+02 2.8064E+02 2.8061E+02

FLAME SPEED = 2.8218E+02 2.6031E+02 1.5519E+02 -7.1045E+05 2.115ME+04 2.0074E+02 2.0036E+02 2.0070E+02 2.0270E+02

FLAME FRONT FROM PH10 = 2.4219E-01 TO PH10 = 4.1627E-01

FLAME FRONT FRUM X = 1.9814E-02 TU X = 6.6422E-02

FLAME THICKNESS = 4.710RE-02 CM

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PO = 1.7695E-01 FLSP = 2.8050E+02 PHNEW = 3.0193E-01

SPN1 = -3.5390E+00 SPN2 = -3.5069E+00 SPTEU1 = -3.5390E+00 DS = 1.0530E-03

I = 5.0000E-01 DT = 1.6201E-02 TOTAL STEPS = 272

NF = 418 NJ = 39

RUN TIME = 3.4065E+01

H

0.4451E-14 1.5346E-10 -2.1723E-10 -7.7438E-10 -1.1824E-09 -1.1027E-09 -1.9479E-10 1.9479E-10 5.4303E-09 3.9667E-09
-3.688AE-09 -1.2143E-08 -1.7213E-08 -1.3491E-08 2.5060E-06 3.1028E-08 9.422AE-08 6.3791E-08 -6.3010E-08 -1.9058E-07
-2.2332E-07 -0.5632E-05 3.7600AE-05 1.2034E-06 2.5060E-06 3.2936E-06 4.1972E-06 6.3043E-06 1.0963E-05 1.9811E-05
J.1926E-05 5.0623E-05 7.6320AE-05 9.9830E-05 1.2674E-04 1.6351E-04 2.0621E-04 2.5149E-04 3.0563E-04 3.6697E-04
0.3509E-06 5.1279E-04 5.9794E-04 6.9152E-04 7.9361E-04 9.0452E-04 1.0243E-03 1.1512E-03 1.2912E-03 1.4399E-03
1.5955E-03 1.1593E-03 1.4263E-03 2.1010E-03 2.2773E-03 2.6449E-03 3.0626E-03 3.4745E-03 3.9459E-03 3.9611E-03
J.2627E-03 3.1266E-03 3.4962E-03 3.5463E-03 3.6830E-03 3.7581E-03 3.8610E-03 3.9495E-03 3.9649E-03 3.5990E-03
J.9622E-03 3.6531E-03 3.9330E-03 3.8931E-03 3.9556E-03 3.1575E-03 3.2144E-03 3.2749E-03 3.3052E-03 2.9443E-03
J.5622E-03 3.6008E-03 3.4024E-03 3.3594E-03 3.2749E-03 3.2144E-03 3.2749E-03 3.2749E-03 3.2749E-03 2.9443E-03
2.9209E-03 2.6633E-03 2.4131E-03 2.7114E-03 2.7130E-03 2.7130E-03 2.7130E-03 2.7130E-03 2.7130E-03 2.7130E-03

UN

1.1025E-10 6.9400E-09 -3.1700E-09 -2.1492E-09 -3.7139E-10 -3.9270E-10 -1.7946E-10 1.7946E-10 1.3540E-07 1.4654E-07
L.1174E-09 -4.1611E-07 -1.4174E-07 -1.4174E-07 -1.4174E-07 -1.4174E-07 -1.4174E-07 -1.4174E-07 -1.4174E-07 -1.4174E-07

1.0227E-06	3.1152E-06	7.7174E-06	1.5638E-05	2.7684E-05	4.5566E-05	7.4510E-05	1.2711E-04	1.8996E-04	2.6512E-04
5.5249E-04	4.5249E-04	6.7869E-04	6.6221E-04	6.4949E-04	6.4949E-04	6.4949E-04	7.1215E-04	7.2911E-04	7.1515E-04
6.9747E-04	6.5249E-04	6.7869E-04	6.6221E-04	6.4949E-04	6.4949E-04	6.4949E-04	6.5435E-04	6.7515E-04	7.0615E-04
7.5462E-04	8.1276E-04	8.1077E-04	9.5694E-04	1.0392E-03	1.1261E-03	1.2157E-03	1.3011E-03	1.3763E-03	1.4603E-03
1.4924E-03	1.5318E-03	1.5576E-03	1.5699E-03	1.5652E-03	1.5652E-03	1.5652E-03	1.5643E-03	1.5119E-03	1.4345E-03
1.3573E-03	1.3264E-03	1.3044E-03	1.2856E-03	1.2666E-03	1.2666E-03	1.2666E-03	1.2666E-03	1.2666E-03	1.2710E-03
1.2633E-03	1.2919E-03	1.2999E-03	1.3069E-03	1.3132E-03	1.3132E-03	1.3132E-03	1.3230E-03	1.3230E-03	1.3230E-03
1.3393E-03	1.3472E-03	1.3557E-03	1.3618E-03	1.3707E-03	1.3707E-03	1.3707E-03	1.3773E-03	1.3824E-03	1.3329E-03
0									
1.0376E-10	1.1736E-08	6.7114E-09	-3.7470E-08	-6.4373E-08	-6.8952E-08	-J.3939E-08	J.7733E-08	2.2353E-07	2.7054E-07
1.0801E-07	6.3168E-06	5.3339E-06	2.3575E-07	7.0769E-07	1.5063E-06	2.9791E-06	3.7466E-06	4.1684E-06	4.9831E-06
6.9344E-06	6.7221E-05	4.6042E-05	4.7221E-05						
2.1168E-04	2.6080E-04	3.1640E-04	3.5763E-04	3.9792E-04	4.3675E-04	4.7356E-04	5.0741E-04	5.3919E-04	5.6691E-04
5.9055E-04	6.1055E-04	6.2969E-04	6.5092E-04	6.7725E-04	7.1164E-04	7.5706E-04	8.0653E-04	8.9298E-04	9.9085E-04
1.1046E-03	1.2604E-03	1.3803E-03	1.5224E-03	1.6607E-03	1.7894E-03	1.9026E-03	1.9683E-03	2.0398E-03	2.0581E-03
2.0470E-03	2.0612E-03	1.9439E-03	1.8444E-03	1.7260E-03	1.5434E-03	1.3696E-03	1.1967E-03	1.0154E-03	8.6192E-03
7.2617E-04	6.1290E-04	5.2643E-04	3.6849E-04	3.6849E-04	3.1830E-04	2.8311E-04	2.5977E-04	2.4514E-04	2.3608E-04
2.2443E-04	2.2103E-04	2.1600E-04	2.0811E-04	2.0314E-04	1.9884E-04	1.9580E-04	1.9150E-04	1.8797E-04	1.8317E-04
1.7696E-04	1.7533E-04	1.7237E-04	1.7002E-04	1.6932E-04	1.6729E-04	1.6694E-04			
H02									
2.1401E-09	-2.1120E-07	-9.3423E-08	2.1863E-07	5.9302E-07	6.5930E-07	9.0807E-07	5.86930E-07	-2.3971E-07	-5.5285E-07
4.4280E-08	1.3715E-06	3.3684E-06	5.8951E-06	8.1031E-06	8.1031E-06	1.2056E-05	1.2056E-05	6.8549E-05	6.8549E-05
1.1486E-04	1.9062E-04	2.0901E-04	4.2223E-04	5.0981E-04	7.2698E-04	8.6568E-04	9.9358E-04	1.0982E-03	1.1671E-03
1.1479E-03	1.1483E-03	1.0557E-03	9.2699E-04	8.2631E-04	7.2909E-04	6.3638E-04	5.5575E-04	4.8245E-04	4.1974E-04
3.6917E-04	3.2976E-04	2.9170E-04	2.0916E-04	2.3366E-04	2.1064E-04	1.9031E-04	1.7270E-04	1.5701E-04	1.4274E-04
1.2948E-06	1.1890E-06	1.1890E-06	1.0730E-04	9.7393E-05	8.1198E-05	7.9589E-05	7.1415E-05	5.6443E-05	4.9724E-05
4.3525E-05	3.7832E-05	3.6646E-05	2.7966E-05	2.3790E-05	1.6729E-05	1.4578E-05	1.1293E-05	8.5898E-06	6.5645E-06
4.9933E-06	3.8321E-06	2.9572E-06	2.0118E-06	1.3172E-06	8.3318E-07	5.1980E-07	3.3701E-07	2.4460E-07	2.0314E-07
1.7200E-07	1.2422E-07	9.2013E-08	7.1245E-08	5.9773E-08	5.4954E-08	5.6143E-08	5.4700E-08	5.3980E-08	5.1073E-08
4.8666E-08	4.6711E-08	4.5190E-08	4.4065E-08	4.3303E-08	4.2869E-08	4.2732E-08			
H202									
2.2049E-10	2.5113E-07	7.6061E-08	-3.2H25E-07	-7.6506E-07	-J.0377E-07	-J.0377E-07	-J.0377E-07	1.0972E-06	1.5931E-06
4.0745E-07	-3.0955E-07	-1.0633E-06	-1.7333E-06	-6.4012E-07	2.5611E-07	6.4097E-06	1.0H74E-05	1.0045E-05	9.5995E-05
1.3136E-05	2.4272E-05	4.6625E-05	8.3M12E-05	1.3942E-04	2.0506E-04	2.9624E-04	4.0H13E-04	5.2R3E-04	6.4969E-04
7.6687E-04	8.5997E-04	9.3099E-04	9.4547E-04	9.3644E-04	9.1360E-04	9.1360E-04	8.2982E-04	7.7195E-04	7.7195E-04
7.6518E-04	6.3088E-04	5.5138E-04	6.6945E-04	3.9789E-04	3.0947E-04	2.3696E-04	1.7293E-04	1.2417E-04	8.1145E-05
5.1669E-05	3.0660E-05	1.7269E-05	5.2345E-05	5.1575E-06	3.4378E-06	2.4752E-06	1.5394E-06	9.2417E-07	5.6651E-07
4.0327E-07	3.7732E-07	4.0755E-07	4.6688E-07	4.3204E-07	3.6900E-07	2.46672E-07	2.41179E-07	2.1080E-07	1.9031E-07
1.7692E-07	1.6121E-07	1.5776E-07	1.3604E-07	1.3604E-07	1.2646E-07	1.2405E-07	1.2405E-07	1.2380E-07	
1.2358E-07	1.2216E-07	1.2200E-07	1.2161E-07	1.2161E-07	1.2065E-07	1.2016E-07	1.1962E-07	1.1899E-07	1.1899E-07
1.1724E-07	1.1654E-07	1.1595E-07	1.1515E-07	1.1495E-07	1.1495E-07	1.1488E-07			
H2									
6.4451E-02	6.4451E-02	6.4451E-02	6.4451E-02	6.4451E-02	6.4451E-02	6.4451E-02	6.4451E-02	6.4451E-02	6.4451E-02
6.4434E-02	6.4434E-02	6.4434E-02	6.4434E-02	6.4434E-02	6.4434E-02	6.4434E-02	6.4434E-02	6.4434E-02	6.4434E-02
6.4577E-02	6.4577E-02	6.4577E-02	6.4577E-02	6.4577E-02	6.4577E-02	6.4577E-02	6.4577E-02	6.4577E-02	6.4577E-02
6.2919E-02	6.2919E-02	6.2919E-02	6.2919E-02	6.2919E-02	6.2919E-02	6.2919E-02	6.2919E-02	6.2919E-02	6.2919E-02
2.8661E-92	5.8059E-92	5.7415E-92	5.6731E-92	5.6004E-92	5.5232E-92	5.4614E-92	5.3734E-92	5.2637E-92	5.1612E-92
4.0657E-02	3.9177E-02	3.9177E-02	3.7659E-02	3.7012E-02	3.5612E-02	3.4744E-02	3.4094E-02	3.3433E-02	3.2321E-02
3.4214E-02	3.4068E-02	3.3965E-02	3.3705E-02	3.3191E-02	3.2503E-02	3.1717E-02	3.0920E-02	3.0127E-02	2.9320E-02
3.4011E-02	3.4068E-02	3.4157E-02	3.4224E-02	3.4291E-02	3.4364E-02	3.4424E-02	3.4484E-02	3.4544E-02	3.4604E-02
3.4710E-02	3.4710E-02	3.4710E-02	3.4710E-02	3.4710E-02	3.4710E-02	3.4710E-02	3.4710E-02	3.4710E-02	3.4710E-02

H2O	6.5015E-08	-1.1076E-07	-1.5378E-06	-2.8976E-06	-3.4570E-06	-2.4827E-06	-7.5970E-07	7.0003E-06	8.7742E-06
He	4.6384E-06	-3.1459E-06	-6.8424E-07	1.0886E-05	3.5319E-05	7.7171E-05	9.4419E-05	9.1072E-05	9.3323E-05
D	2.3208E-06	-1.7808E-04	3.4197E-04	6.7513E-04	1.0932E-03	1.4522E-03	1.9902E-03	2.6176E-03	3.6047E-03
T	1.2269E-06	-2.1780E-04	3.4197E-04	6.7513E-04	1.0932E-03	1.4522E-03	1.9902E-03	2.6176E-03	3.6047E-03
Li	4.6384E-06	-3.1459E-06	-6.8424E-07	1.0886E-05	3.5319E-05	7.7171E-05	9.4419E-05	9.1072E-05	9.3323E-05
Be	2.3208E-06	-1.7808E-04	3.4197E-04	6.7513E-04	1.0932E-03	1.4522E-03	1.9902E-03	2.6176E-03	3.6047E-03
B	1.2269E-06	-2.1780E-04	3.4197E-04	6.7513E-04	1.0932E-03	1.4522E-03	1.9902E-03	2.6176E-03	3.6047E-03
C	4.6384E-03	1.0915E-02	1.2993E-02	1.5336E-02	1.7946E-02	2.0832E-02	2.3955E-02	2.7373E-02	3.1047E-02
N	2.3208E-03	4.3676E-02	4.8459E-02	5.3561E-02	5.9024E-02	6.4852E-02	7.1017E-02	7.7716E-02	8.4554E-02
O	1.2269E-03	3.9191E-02	3.9497E-02	3.9549E-02	3.9598E-02	1.3598E-01	1.4522E-01	1.5437E-01	1.7146E-01
F	4.6384E-02	1.0907E-01	1.1763E-01	1.2883E-01	1.3598E-01	2.0761E-01	2.1499E-01	2.2096E-01	2.2594E-01
Ne	2.3208E-02	1.0906E-01	1.0907E-01	1.0915E-01	1.0924E-01	2.4026E-01	2.4626E-01	2.4949E-01	2.5319E-01
Mg	1.2269E-02	1.0905E-01	1.0907E-01	1.0915E-01	1.0924E-01	2.4026E-01	2.4626E-01	2.4949E-01	2.5319E-01
Si	4.6384E-01	2.3735E-01	2.3876E-01	2.4026E-01	2.4342E-01	2.4626E-01	2.4949E-01	2.5319E-01	2.5633E-01
P	2.3208E-01	2.2433E-01	2.2434E-01	2.2435E-01	2.2436E-01	2.2437E-01	2.2438E-01	2.2439E-01	2.2440E-01
S	1.2269E-01	2.2432E-01	2.2433E-01	2.2434E-01	2.2435E-01	2.2436E-01	2.2437E-01	2.2438E-01	2.2439E-01
Cl	4.6384E-01	2.2431E-01	2.2432E-01	2.2433E-01	2.2434E-01	2.2435E-01	2.2436E-01	2.2437E-01	2.2438E-01
Ar	2.3208E-01	2.2430E-01	2.2431E-01	2.2432E-01	2.2433E-01	2.2434E-01	2.2435E-01	2.2436E-01	2.2437E-01
K	1.2269E-01	2.2429E-01	2.2430E-01	2.2431E-01	2.2432E-01	2.2433E-01	2.2434E-01	2.2435E-01	2.2436E-01
Ca	4.6384E-01	2.2428E-01	2.2429E-01	2.2430E-01	2.2431E-01	2.2432E-01	2.2433E-01	2.2434E-01	2.2435E-01
Sc	2.3208E-01	2.2427E-01	2.2428E-01	2.2429E-01	2.2430E-01	2.2431E-01	2.2432E-01	2.2433E-01	2.2434E-01
Ti	1.2269E-01	2.2426E-01	2.2427E-01	2.2428E-01	2.2429E-01	2.2430E-01	2.2431E-01	2.2432E-01	2.2433E-01
V	4.6384E-01	2.2425E-01	2.2426E-01	2.2427E-01	2.2428E-01	2.2429E-01	2.2430E-01	2.2431E-01	2.2432E-01
Cr	2.3208E-01	2.2424E-01	2.2425E-01	2.2426E-01	2.2427E-01	2.2428E-01	2.2429E-01	2.2430E-01	2.2431E-01
Mn	1.2269E-01	2.2423E-01	2.2424E-01	2.2425E-01	2.2426E-01	2.2427E-01	2.2428E-01	2.2429E-01	2.2430E-01
Fe	4.6384E-01	2.2422E-01	2.2423E-01	2.2424E-01	2.2425E-01	2.2426E-01	2.2427E-01	2.2428E-01	2.2429E-01
Co	2.3208E-01	2.2421E-01	2.2422E-01	2.2423E-01	2.2424E-01	2.2425E-01	2.2426E-01	2.2427E-01	2.2428E-01
Ni	1.2269E-01	2.2420E-01	2.2421E-01	2.2422E-01	2.2423E-01	2.2424E-01	2.2425E-01	2.2426E-01	2.2427E-01
Cu	4.6384E-01	2.2419E-01	2.2420E-01	2.2421E-01	2.2422E-01	2.2423E-01	2.2424E-01	2.2425E-01	2.2426E-01
Zn	2.3208E-01	2.2418E-01	2.2419E-01	2.2420E-01	2.2421E-01	2.2422E-01	2.2423E-01	2.2424E-01	2.2425E-01
Ga	1.2269E-01	2.2417E-01	2.2418E-01	2.2419E-01	2.2420E-01	2.2421E-01	2.2422E-01	2.2423E-01	2.2424E-01
In	4.6384E-01	2.2416E-01	2.2417E-01	2.2418E-01	2.2419E-01	2.2420E-01	2.2421E-01	2.2422E-01	2.2423E-01
Tl	2.3208E-01	2.2415E-01	2.2416E-01	2.2417E-01	2.2418E-01	2.2419E-01	2.2420E-01	2.2421E-01	2.2422E-01
Pb	1.2269E-01	2.2414E-01	2.2415E-01	2.2416E-01	2.2417E-01	2.2418E-01	2.2419E-01	2.2420E-01	2.2421E-01
Bi	4.6384E-01	2.2413E-01	2.2414E-01	2.2415E-01	2.2416E-01	2.2417E-01	2.2418E-01	2.2419E-01	2.2420E-01
Pr	2.3208E-01	2.2412E-01	2.2413E-01	2.2414E-01	2.2415E-01	2.2416E-01	2.2417E-01	2.2418E-01	2.2419E-01
Ta	1.2269E-01	2.2411E-01	2.2412E-01	2.2413E-01	2.2414E-01	2.2415E-01	2.2416E-01	2.2417E-01	2.2418E-01
Hf	4.6384E-01	2.2410E-01	2.2411E-01	2.2412E-01	2.2413E-01	2.2414E-01	2.2415E-01	2.2416E-01	2.2417E-01
Ru	2.3208E-01	2.2409E-01	2.2410E-01	2.2411E-01	2.2412E-01	2.2413E-01	2.2414E-01	2.2415E-01	2.2416E-01
Rh	1.2269E-01	2.2408E-01	2.2409E-01	2.2410E-01	2.2411E-01	2.2412E-01	2.2413E-01	2.2414E-01	2.2415E-01
Pt	4.6384E-01	2.2407E-01	2.2408E-01	2.2409E-01	2.2410E-01	2.2411E-01	2.2412E-01	2.2413E-01	2.2414E-01
Au	2.3208E-01	2.2406E-01	2.2407E-01	2.2408E-01	2.2409E-01	2.2410E-01	2.2411E-01	2.2412E-01	2.2413E-01
Ag	1.2269E-01	2.2405E-01	2.2406E-01	2.2407E-01	2.2408E-01	2.2409E-01	2.2410E-01	2.2411E-01	2.2412E-01
Os	4.6384E-01	2.2404E-01	2.2405E-01	2.2406E-01	2.2407E-01	2.2408E-01	2.2409E-01	2.2410E-01	2.2411E-01
Ir	2.3208E-01	2.2403E-01	2.2404E-01	2.2405E-01	2.2406E-01	2.2407E-01	2.2408E-01	2.2409E-01	2.2410E-01
Ru	1.2269E-01	2.2402E-01	2.2403E-01	2.2404E-01	2.2405E-01	2.2406E-01	2.2407E-01	2.2408E-01	2.2409E-01
Pd	4.6384E-01	2.2401E-01	2.2402E-01	2.2403E-01	2.2404E-01	2.2405E-01	2.2406E-01	2.2407E-01	2.2408E-01
Ag	2.3208E-01	2.2400E-01	2.2401E-01	2.2402E-01	2.2403E-01	2.2404E-01	2.2405E-01	2.2406E-01	2.2407E-01
Al	1.2269E-01	2.2399E-01	2.2400E-01	2.2401E-01	2.2402E-01	2.2403E-01	2.2404E-01	2.2405E-01	2.2406E-01
Sc	4.6384E-01	2.2398E-01	2.2399E-01	2.2400E-01	2.2401E-01	2.2402E-01	2.2403E-01	2.2404E-01	2.2405E-01
Y	2.3208E-01	2.2397E-01	2.2398E-01	2.2399E-01	2.2400E-01	2.2401E-01	2.2402E-01	2.2403E-01	2.2404E-01
La	1.2269E-01	2.2396E-01	2.2397E-01	2.2398E-01	2.2399E-01	2.2400E-01	2.2401E-01	2.2402E-01	2.2403E-01
Ce	4.6384E-01	2.2395E-01	2.2396E-01	2.2397E-01	2.2398E-01	2.2399E-01	2.2400E-01	2.2401E-01	2.2402E-01
Pr	2.3208E-01	2.2394E-01	2.2395E-01	2.2396E-01	2.2397E-01	2.2398E-01	2.2399E-01	2.2400E-01	2.2401E-01
Nd	1.2269E-01	2.2393E-01	2.2394E-01	2.2395E-01	2.2396E-01	2.2397E-01	2.2398E-01	2.2399E-01	2.2400E-01
Eu	4.6384E-01	2.2392E-01	2.2393E-01	2.2394E-01	2.2395E-01	2.2396E-01	2.2397E-01	2.2398E-01	2.2399E-01
Gd	2.3208E-01	2.2391E-01	2.2392E-01	2.2393E-01	2.2394E-01	2.2395E-01	2.2396E-01	2.2397E-01	2.2398E-01
Tb	1.2269E-01	2.2390E-01	2.2391E-01	2.2392E-01	2.2393E-01	2.2394E-01	2.2395E-01	2.2396E-01	2.2397E-01
Dy	4.6384E-01	2.2389E-01	2.2390E-01	2.2391E-01	2.2392E-01	2.2393E-01	2.2394E-01	2.2395E-01	2.2396E-01
Ho	2.3208E-01	2.2388E-01	2.2389E-01	2.2390E-01	2.2391E-01	2.2392E-01	2.2393E-01	2.2394E-01	2.2395E-01
Er	1.2269E-01	2.2387E-01	2.2388E-01	2.2389E-01	2.2390E-01	2.2391E-01	2.2392E-01	2.2393E-01	2.2394E-01
Tm	4.6384E-01	2.2386E-01	2.2387E-01	2.2388E-01	2.2389E-01	2.2390E-01	2.2391E-01	2.2392E-01	2.2393E-01
Yb	2.3208E-01	2.2385E-01	2.2386E-01	2.2387E-01	2.2388E-01	2.2389E-01	2.2390E-01	2.2391E-01	2.2392E-01
Lu	1.2269E-01	2.2384E-01	2.2385E-01	2.2386E-01	2.2387E-01	2.2388E-01	2.2389E-01	2.2390E-01	2.2391E-01
Yttrium	4.6384E-01	2.2383E-01	2.2384E-01	2.2385E-01	2.2386E-01	2.2387E-01	2.2388E-01	2.2389E-01	2.2390E-01
Ytterbium	2.3208E-01	2.2382E-01	2.2383E-01	2.2384E-01	2.2385E-01	2.2386E-01	2.2387E-01	2.2388E-01	2.2389E-01
Yttrium-Iridium-Garnet	1.2269E-01	2.2381E-01	2.2382E-01	2.2383E-01	2.2384E-01	2.2385E-01	2.2386E-01	2.2387E-01	2.2388E-01

X IN CM

Y	1.461E-03	9.714E-04	1.942E-03	2.914E-03	3.865E-03	4.856E-03	5.828E-03	6.749E-03	7.714E-03	H.4587E-03
Z	1.461E-03	9.833E-04	1.0521E-02	1.1201E-02	1.1897E-02	1.2587E-02	1.3219E-02	1.3764E-02	1.4262E-02	1.4756E-02
UH	1.5552E-02	1.5754E-02	1.6262E-02	1.6779E-02	1.7304E-02	1.7844E-02	1.8409E-02	1.8944E-02	1.9361E-02	1.9756E-02
D	1.9830E-02	2.0320E-02	2.0839E-02	2.1224E-02	2.1628E-02	2.2054E-02	2.2491E-02	2.2933E-02	2.3435E-02	2.3940E-02
MU2	2.4466E-02	2.5016E-02	2.5589E-02	2.6116E-02	2.6607E-02	2.7145E-02	2.7612E-02	2.8115E-02	2.8633E-02	3.0276E-02
H2	3.1042E-02	3.1831E-02	3.2643E-02	3.3474E-02	3.4334E-02	3.5212E-02	3.6108E-02	3.7044E-02	3.7957E-02	3.8907E-02
U2	3.9872E-02	4.0851E-02	4.2043E-02	4.2947E-02	4.3861E-02	4.5313E-02	4.6782E-02	4.8257E-02	4.9765E-02	5.1274E-02
M20	5.2794E-02	5.4324E-02	5.5861E-02	5.8049E-02	5.8049E-02	6.0252E-02	6.2457E-02	6.4693E-02	6.9114E-02	7.1644E-02
M2	7.3714E-02	7.6960E-02	8.0183E-02	8.3443E-02	8.6720E-02	9.0112E-02	9.3319E-02	9.6639E-02	9.9970E-02	1.0470E-02
MD	1.0594E-01	1.1425E-01	1.1906E-01	1.2390E-01	1.2875E-01	1.3362E-01	1.3850E-01	1.4350E-01	1.4850E-01	1.5350E-01

FLAME FRONT FROM PHI0 = 2.4219E-01 TU PHI0 = 4.1627E-01

FLAME FRONT FROM K = 1.9411E-02 TU K = 6.6124E-02

FLAME THICKNESS = 4.7110E-02 CM

9	97	1.00000000F+00	5.2103171E-04	9.08674678E-03	2.70604230E-01	5.58288874E+02	2.7426008E-03	3.7226642E+03	1.8196720ME-03	1.0941928E-03
H	1.70000000E+01	5.58288874E+02	2.00000000E+01	3.7226642E+03	2.7426008E-03	1.5456505E+02	1.8196720ME-03	1.0941928E-03	1.0941928E-03	1.0941928E-03
UH	1.60000000E+01	3.7226642E+03	1.70000000E+01	1.8196720ME-03	1.0941928E-03	1.0941928E-03	1.0941928E-03	1.0941928E-03	1.0941928E-03	1.0941928E-03
D	3.40000000E+01	9.56829781E+02	1.00000000E+01	1.0941928E-03	1.0941928E-03	1.0941928E-03	1.0941928E-03	1.0941928E-03	1.0941928E-03	1.0941928E-03
MU2	2.00000000E+00	-1.59507953E-01	4.60099148E-01	4.60099148E-01	4.60099148E-01	3.20000000E+00	-3.56689080E-02	1.88488399E-03	2.36766026E-03	1.88488399E-03
H2	3.20000000E+00	-3.56689080E-02	1.88488399E-03	2.36766026E-03	1.88488399E-03	2.80000000E+01	-3.951632AE-02	0.	0.	0.
U2	3.20000000E+00	-3.21097059E+03	2.36766026E-03	1.88488399E-03	1.88488399E-03	2.80000000E+01	-3.951632AE-02	0.	0.	0.
M20	1.60000000E+01	-3.951632AE-02	2.80000000E+01	0.	0.	0.	0.	0.	0.	0.
M2	2.80000000E+01	0.	0.	0.	0.	0.	0.	0.	0.	0.

MD	1.7698E-01	F1SP = 2.8063E+02	PHI0 = 3.0193E-01	MD = 1.7698E-01	F1SP = 2.8063E+02	PHI0 = 3.0193E-01	MD = 1.7698E-01	F1SP = 2.8063E+02	PHI0 = 3.0193E-01	MD = 1.7698E-01
2.00000000E+00 5.00000000E+03 1.00000000E+03										

2.00000000E+00 5.00000000E+03 1.00000000E+03

2.98562202E-01 6.5563274E-02 2.98000000E+02

1.00000000E+00 1.2187694E-02

CONDUCTION POINTS

76	2.071366E-02	7.730441E-02	1.126648E-01	1.526915E-01	1.774841E-01	2.054795E-01	2.36597E-01	3.29629E-01	3.94861E-01	4.19361E-01
0	2.436670E-01	2.561694E-01	2.703171E-01	2.806739E-01	2.948210E-01	3.051784E-01	3.193261E-01	3.22659AE-01	3.22659AE-01	3.22659AE-01
3.03H30562-01	3.563324E-01	3.763403E-01	3.944205E-01	4.223153E-01	4.473195E-01	4.673195E-01	4.873195E-01	4.973195E-01	5.073195E-01	5.22659AE-01
5.742963E-01	6.009000E-01	6.6655AE-01	7.04216E-10	-H.303130E-10	1.109424E-08	1.923710E-08	2.889144E-07	3.436122E-06	3.436122E-06	3.436122E-06
H.4950H4E-14	H.6655AE-01	H.6655AE-01	3.23811AE-04	5.707617E-04	1.063220E-03	1.5466234E-03	2.342472E-03	2.901680E-03	2.901680E-03	2.901680E-03
3.703242E-05	1.19193E-04	3.444172E-03	3.959KH1E-03	3.669946E-03	1.632024M-03	3.421251E-03	3.140542E-03	2.93946AE-03	2.93946AE-03	2.93946AE-03
3.522870E-03	3.444172E-03	2.713920E-03	-5.073133E-04	4.935329E-04	3.033119AE-04	1.34434J9E-04	4.777772HE-06	6.744663E-05	6.744663E-05	6.744663E-05
C.733145E-03	2.713920E-03	H.217861E-10	6.667579E-04	7.247153E-04	6.674354E-04	3.4272714E-03	1.2974734E-03	1.337156E-03	1.337156E-03	1.337156E-03
J.8106662E-04	6.667579E-04	1.522364E-03	1.366901E-03	1.276444E-03						
1.542673E-03	1.542673E-03	1.046160E-10	-1.204945E-04	6.0855170E-04	6.0855170E-04	1.730361E-04	1.070417E-03	1.701461E-03	2.027629E-03	2.027629E-03
1.372631E-03	1.372631E-03	1.046160E-10	-1.204945E-04	6.0855170E-04	6.0855170E-04	1.730361E-04	1.070417E-03	1.701461E-03	2.027629E-03	2.027629E-03
1.037616E-10	1.037616E-10	1.037616E-10	-1.204945E-04	6.0855170E-04	6.0855170E-04	1.730361E-04	1.070417E-03	1.701461E-03	2.027629E-03	2.027629E-03
1.911591E-03	1.669331E-03	1.669331E-03	6.0855170E-04	1.037616E-10						
1.67929E-04	1.669331E-04	1.669331E-04	6.0855170E-04	1.037616E-10						
/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111	/1.4671111111111111

1.14272E-03	8.57369E-04	4.618629E-04	3.0224943E-04	1.046216E-04	1.337297E-04	8.547893E-05	5.662451E-05
3.11644E-05	1.57717E-05	4.597169E-05	1.507743E-05	2.283767E-05	1.007324E-05	5.4792H7E-05	4.935581E-05
4.31349E-08	4.27324E-08	1.63H916E-07	-8.181571E-07	1.-c066643E-06	1.-730431E-06	1.042506E-05	5.637645E-05
c.20492E-10	1.63H916E-07	8.12940E-04	8.12940E-04	2.162042E-04	2.162042E-04	5.97272E-05	2.651095E-04
7.91415E-04	9.45580E-04	1.736816E-07	1.384435E-07	1.239794E-07	1.239794E-07	1.223490E-07	1.0H4060E-06
4.22723E-07	3.039779E-07	1.150745E-07	1.148776E-07	1.148776E-07	1.148776E-07	1.200039E-07	1.174037E-07
6.485084E-02	6.485024E-02	6.485024E-02	6.485024E-02	6.483671E-02	6.483671E-02	6.483671E-02	6.390176E-02
6.281797E-02	6.101351E-02	5.959232E-02	5.959232E-02	5.61H7E-02	5.415119E-02	5.09753E-02	4.227400E-02
3.81613E-02	3.57944E-02	3.416574E-02	3.416574E-02	3.383061E-02	3.391701E-02	3.413461E-02	3.46H553E-02
3.449793E-02	3.5010996E-02	2.179023E-01	2.179023E-01	2.178R69E-01	2.178125E-01	2.178125E-01	2.15H923E-01
c.17A994E-01	2.178982E-01	1.938710E-01	1.938710E-01	1.628H8E-01	1.428174E-01	1.007349E-01	7.756356E-02
c.111397E-01	2.0493H4E-01	2.192041E-01	2.192041E-01	2.037501E-03	3.529965E-04	1.604470E-04	8.990373E-05
4.280195E-02	2.210425E-02	6.639131E-03	2.210425E-02	7.584760E-05	1.718319E-05	4.667241E-04	1.H16660E-03
1.639545E-05	2.301260E-07	-1.7561H9E-06	6.679335E-06	1.0718319E-05	9.2669714E-05	4.667241E-04	1.H16660E-03
6.992510E-03	1.439776E-02	2.948159E-02	4.25573E-02	1.673452E-02	9.008757L-02	1.2495462E-01	1.605500E-01
1.971313E-01	2.192041E-01	2.361270E-01	2.410771E-01	2.431674E-01	2.4333655E-01	2.434373E-01	
c.36170E-01	2.434128E-01	2.979426E-01	2.980245E-01	2.992196E-01	3.024931E-01	3.167305E-01	3.46H703E-01
c.980000E-01	2.979426E-01	6.011881E-01	6.275526E-01	7.28748E-01	7.73472E-01	9.879669E-01	1.220536E+00
4.251999E-01	1.372663E+00	1.430418E+00	1.463114E+00	1.501979E+00	1.529452E+00	1.592574E+00	
1.626770E+00	1.632729E+00	2.749240E-03	2.749240E-03	1.819672E-03	1.809419E-03	4.600991E-03	2.367660E-03
9.04674E-03	2.706642E-03	u.					

INDEX = 0

GLOSSARY

- c_{pk} = specific heat of species k, cal-g⁻¹ - K⁻¹.
 c_p = specific heat of the mixture, cal-g⁻¹ - K⁻¹.
 D_{km} = diffusion coefficient of species k in the mixture, cm²-s⁻¹.
 h_k = specific enthalpy of species k, cal-g⁻¹.
 k_j = rate constant for reaction j in centimeter - mole - second units.
 m_0 = mass flux of the mixture through the origin, g-cm⁻²-s⁻¹.
 M_k = molecular weight of species k, g-mole⁻¹.
 N = number of chemical species (also number of PDE's).
 NO = number of ODE's.
 p = pressure, atmos.
 r_j = rate of reaction j, mole-cm⁻¹-s⁻¹.
 R = gas constant = 82.05 cm³-atoms-mole⁻¹-K⁻¹.
 R_k = rate of production of species k, mole-cm⁻³-s⁻¹.
 S_v = velocity of the flame relative to the unburned mixture, cm-s⁻¹.
 \hat{t} = temporal coordinate, s. (\hat{t} , \hat{x} coordinate system).
 \tilde{t} = temporal coordinate, s. (\tilde{t} , $\tilde{\psi}$ coordinate system).
 t = non dimensional temporal coordinate, with respect to t_∞ (t , ψ coordinate system).
 \hat{T} = temperature of the mixture, K.
 \tilde{T} = temperature of the mixture, K.
 T = non dimensional temperature of the mixture, with respect to T_∞ .
 T_U = temperature of the unburned mixture.

GLOSSARY (continued)

- T_B = temperature of the burned mixture (adiabatic temperature).
 v = fluid velocity of the mixture, $\text{cm}\cdot\text{s}^{-1}$.
 V_k = diffusion velocity of species k , $\text{cm}\cdot\text{s}^{-1}$.
 \hat{x} = spatial coordinate, cm.
 Y_k = mass fraction of species k .
 Y_{kU} = mass fraction of species k in the unburned mixture.
 Y_{kB} = mass fraction of species k in the burned mixture.
 ϵ_k = mass flux fraction of species k .
 λ = heat conductivity of the mixture, $\text{cal}\cdot\text{cm}^{-1}\cdot\text{s}^{-1}\cdot\text{K}^{-1}$.
 ρ = density of the mixture, $\text{g}\cdot\text{cm}^{-3}$.
 $\tilde{\psi}$ = transformed distance coordinate, $\text{g}\cdot\text{cm}^{-2}$.
 ψ = non dimensional transformed distance coordinate, with respect to ψ_∞ .

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