

LEVEL

(D)

AD A106352

THE ARL/FED ANECHOIC CHAMBER

R. C. Marboe and J. M. Fitzgerald

Technical Memorandum
File No. TM 81-164
4 August, 1981
Contract No. N00024-81-6043 *NEW*

**DTIC
ELECTE
OCT 28 1981
S H D**

Copy No. 43

The Pennsylvania State University
APPLIED RESEARCH LABORATORY
Post Office Box 30
State College, PA 16801

Approved for Public Release
Distribution Unlimited

NAVY DEPARTMENT

NAVAL SEA SYSTEMS COMMAND

DTIC FILE COPY

81 10 21

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER AR-100/TM-81-164	2. GOVT ACCESSION NO. AD-A106 352	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) THE ARL/FED ANECHOIC CHAMBER		5. TYPE OF REPORT & PERIOD COVERED Technical Memorandum
7. AUTHOR(s) R. C. Marboe and J. M. Fitzgerald		8. CONTRACT OR GRANT NUMBER(s) N00024-81-6043
9. PERFORMING ORGANIZATION NAME AND ADDRESS Applied Research Laboratory Post Office Box 30 State College, Pa 16801		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
11. CONTROLLING OFFICE NAME AND ADDRESS		12. REPORT DATE 11/4 Aug 1981
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		13. NUMBER OF PAGES 25
		15. SECURITY CLASS. (of this report) Unclassified
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release. Distribution unlimited Per NAVSEA - Oct. 1, 1981.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENT. BY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) anechoic, chamber, design, characteristics		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) <p>This report documents the design and acoustic characteristics of the Applied Research Laboratory Department anechoic chamber. It is located on the above-console deck of the Garfield Thomas Water Tunnel situated so as to allow its use in conjunction with the Axial Flow Research Fan (AFRF). The internal free dimensions of the chamber are 2.74 x 3.05 x 1.98 meters (9 x 10 x 6.5 ft.). The sound absorbing walls are composed of polyurethane foam acoustic wedges, air voids, embossed</p>		

DD FORM 1473 EDITION OF 1 NOV 65 IS OBSOLETE
1 JAN 73

391007

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

49

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

acoustic foam wall lining, plywood and wood frame members. Air intake ducts, baffled and lined with a lead-foam sheet, provide an adequate air volume for the AFRF. Based on measurements of transmission loss and inverse square law, the chamber is considered anechoic for frequencies above 230 Hz and semi-anechoic for lower frequencies.

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

Accession For	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By _____	
Distribution/	
Availability Codes	
Dist	Special
A	

Subject: The ARL/FED Anechoic Chamber

References: See page 9

Abstract: This report documents the design and acoustic characteristics of the Applied Research Laboratory Department anechoic chamber. It is located on the above-console deck of the Garfield Thomas Water Tunnel situated so as to allow its use in conjunction with the Axial Flow Research Fan (AFRF). The internal free dimensions of the chamber are 2.74 x 3.05 x 1.98 meters (9 x 10 x 6.5 ft.). The sound absorbing walls are composed of polyurethane foam acoustic wedges, air voids, embossed acoustic foam wall lining, plywood and wood frame members. Air intake ducts, baffled and lined with a lead-foam sheet, provide an adequate air volume for the AFRF. Based on measurements of transmission loss and inverse-square law, the chamber is considered anechoic for frequencies above 230 Hz and semi-anechoic for lower frequencies.

Acknowledgment: This work has been supported by the Naval Sea Systems Command Codes 05H and 63R31.

TABLE OF CONTENTS

Abstract 1

Table of Contents 2

List of Figures 3

I. INTRODUCTION 4

II. DESIGN 4

III. CHAMBER SPECIFICATIONS 4

 A. Geometry 4

 B. Air Inlets 5

 C. Use With and Without Fan Ducts 5

IV. CALIBRATION 5

 A. Ambient Noise 5

 B. Inverse Square Law 6

 Results 6

 C. Transmission Loss 7

 Results 7

 D. Directivity 7

V. CONCLUSIONS 8

VI. REFERENCES 9

 Figures 10

LIST OF FIGURES

<u>Figure</u>	<u>Title</u>	<u>Page</u>
1	Location of Anechoic Chamber on Above-Console Deck	10
2	Front View of Anechoic Chamber	11
3	Fan Inlet Wall	12
4	Typical Cross Section of Chamber Wall and Ceiling Sections	13
5	Typical Cross Section of Intake Duct	14
6	Ambient Noise Sound Pressure Spectrum	15
7	Instrumentation Schematic for Inverse Square Law Measurement	16
8	Inverse Square Law Measurement for Selected Tones (Without Fan Duct)	17
9	Instrumentation Schematic for Transmission Loss Measurement	18
10	Transmission Loss for Side Wall (Away From Door)	19
11	Transmission Loss Through the Door	20
12	Transmission Loss for Fan Inlet Wall (Fan Duct Closed)	21
13	Transmission Loss for Air Intake Duct (3 Feet From End)	22
14	Instrumentation Schematic for Directivity Measurement	23
15	Tonal Directivity Patterns at 100 Hz, 150 Hz and 200 Hz (Source Positioned at Fan Inlet)	24
16	Tonal Directivity Patterns at 250 Hz and 300 Hz (Source Positioned at Fan Inlet)	25

I. INTRODUCTION

The need for an anechoic chamber with a cut-off frequency low enough to allow rotor noise studies resulted in the redesign of the former ARL/FED semi-anechoic chamber [1].* The chamber described in this report was subsequently constructed and acoustically calibrated. The chamber is located on the above-console deck of the Garfield Thomas Water Tunnel in a position where it may be used with or without the Axial Flow Research Fan (AFRF). Figure 1 shows this location on the deck. The free internal dimensions are 2.74 m (9 ft.) wide x 3.05 m (10 ft.) long x 1.98 m (6.5 ft.) high. Acoustically treated air ducts supply an adequate volume of air for the AFRF. The chamber could be employed for rotor studies, inlet configuration studies, microphone/speaker calibration or other acoustic programs requiring at least a semi-anechoic chamber.

II. DESIGN

The chamber was designed for an overall cut-off frequency of 250 Hz. The polyurethane foam acoustic wedges were commercially prepared using dimension specifications obtained from Beranek and Sleeper [2]. The dimensions chosen correspond to a wedge anechoic cut-off frequency of 170 Hz. This is under the required 250 Hz frequency and provides a larger sound absorption area. Though some wedges required some modification, their dimensions still satisfy the 250 Hz dimensions.

The major design requirement which sets this chamber apart from the previous one is the necessity of ducting air into the chamber for the AFRF. Baffled and lined compact ducts were designed to attenuate 25 dB yet result in as low a flow resistance as possible.

III. CHAMBER SPECIFICATIONS

A. Geometry

The existing wooden wall, ceiling, base and floor frames from the previous semi-anechoic chamber were reused. They are specified by Lauchle and Wong [1].

The chamber rests atop a support platform built to raise the chamber to the centerline of the AFRF. This platform is filled with a plastic encased 20 cm layer of fiberglass recovered from the previous chamber. The wooden framework base, measuring 3.96 m (13 ft.) wide x 4.27 m (14 ft.) long, is bolted to 12 commercial kinetic isolators which sit atop the support platform. The spaces in the floor directly above the base are covered with a 25 mm (1") lead-foam liner beneath a turkey wire floor cover. This provides necessary sound and vibration attenuation from beneath the chamber. Figures 2 and 3 give views of the entire chamber.

The wall and ceiling construction consists of plywood, acoustic foam liner and commercially prepared foam wedges. This is shown in Figure 4. The floor of the chamber has wedges mounted on removable

* Numbers in brackets denote references at the end of the report.

19 mm (3/4 in.) plywood sheets which rest on the turkey wire covered floor. These can be lifted and removed from the chamber to work on test equipment. Several single wedges lift off for single stepping points.

For directivity measurements, a permanent microphone boom is installed. A stepping motor allows 1.8° angular increments for surveys. The maximum boom radius for a full 180° survey is 1.37 m (4.5 ft.) from the center of the inlet.

It should be noted that the height of the chamber exceeds the height of the overhead crane rail; therefore, new crane stops were installed previous to the chamber. This reduces the range of the crane by about 4.5 m (15 ft.) A 1.7 m (5.5 ft.) x 2.1 m (7 ft.) walk platform was also built at the level of the door. The door may be rolled into and out of the door frame and rested on the walk platform when not in use.

B. Air Inlets

Operation of the AFRF in conjunction with the anechoic chamber necessitated construction of air passages into the chamber. This was accomplished by raising the walls 51 mm (2 in.) off the base and the ceiling another 51 mm (2 in.) off the wall. A duct arrangement was added around the top and on three sides of the bottom with a plywood cover on the bottom of the fan inlet wall. These ducts were baffled and lined with a foam-lead-foam sound barrier. A typical cross section of the intake duct is shown in Figure 5.

C. Use With and Without Fan Ducts

Mounting of test ducts is done through the fan inlet hole. The AFRF fits through the hole and the bellmouth can be mounted on it from the inside of the chamber. Other types of ducts can be mounted in the fan inlet hole if their diameter is less than 63 cm (25 in.) In this case, the AFRF needs to have a section removed and the entire unit rolled back. For each duct configuration or when the fan inlet is covered, wedges can be mounted to fill the wall. Each wedge around the fan inlet is individually mounted with velcro.

IV. CALIBRATION

A. Ambient Noise

Ambient (or background) noise levels were measured in the anechoic chamber without the operation of the water tunnel test facilities or ancillary equipment. Therefore this represents the most quiet background condition that may be achieved.

Figure 6 illustrates the narrow band spectral results obtained. Overall, the ambient noise levels are very low although significant low frequency sound (i.e. below 20 Hz) exists within the chamber. Additional peaks are apparent at the line frequency

(60 Hz) and its first harmonic (120 Hz). The peaks observed at 395 Hz and 800 Hz are most likely harmonically related although their origin is unknown.

It is recommended that high pass filtering be used for the measurement of low level sound sources within the chamber. This will prevent signal overload caused by the low frequency sound in the chamber. Improved cable shielding and placement will help to reduce line frequency noise.

B. Inverse Square Law

In an ideal free-field, the near-field sound intensity decreases with the fourth power of separation distance between source and receiver. As this separation is further increased, the decrease in sound intensity obeys an inverse-square law which signifies the far-field condition. Skudrzyk [3] describes this result in detail for a simple source.

To verify the free-field properties of the anechoic chamber, measurements of the inverse-square law were performed. Figure 7 illustrates the test set-up employed. All measurements were performed along a horizontal diagonal of the chamber, 2.0 m (6.6 ft.) above the floor. One set of measurements was conducted with the loudspeaker source one foot off-center and another set with the source five feet off center. The microphone was traversed in one-foot increments along the diagonal on the speaker axis. The loudspeaker was driven with pure tones at constant voltage from 100 Hz to 15 kHz. The microphone was calibrated with a B & K type 4220 Pistonphone and calibration was checked at the conclusion of the measurements (no discrepancies were found).

Results

The results of the inverse square law measurements are summarized in Figure 8. A 6 dB drop in sound pressure level per distance doubled of separation of the source and microphone indicates a good free-field condition. Figure 8 indicates such a condition for frequencies above 200 Hz. Scatter at lower frequencies indicate a semi-anechoic condition due to the presence of standing waves. Placement of the source near a corner does not substantially alter the free-field simulation above 200 Hz. It should be noted in Figure 8 that when the microphone is within 30 cm (1 ft.) of the wall, a wall interaction effect is shown. This limits the volume of the free-field to within a quarter wavelength of the chamber walls. Also, directivity measurements are limited to 1.37 m (4.5 ft.) from source for $\pm 90^\circ$ from axis and 1.95 m (6.4 ft.) from source for $\pm 45^\circ$ from axis.

C. Transmission Loss

The transmission loss is a measure of the amount of noise attenuation provided by a sound insulating barrier. This attenuation, as a function of frequency, was measured at four positions using the test set-up illustrated in Figure 9. The measurement positions were:

1. Side Wall - .91 m (3 ft.) from an end, through the wall
2. Side Wall - .91 m (3 ft.) from an end, through the door
3. End Wall - .91 m (3 ft.) from an end, through the fan inlet wall with the fan duct closed
4. Side Wall - .91 m (3 ft.) from an end with the source and external microphone vertically positioned at the duct inlet, through the duct

The loudspeaker was driven by pure tones, at constant voltage, ranging from 40 Hz to 1 kHz. Both microphones were calibrated using a B & K type 4220 Pistonphone and their levels were equalized using the input attenuators on the Spectral Dynamics 360 analyzer. Use of the two channel feature of the SD 360 permitted direct measurement of the transmission loss.

Results

The transmission loss as a function of frequency is shown in Figures 10 through 13. Figures 10 and 11 show the attenuation for a typical wall with 12.7 mm (1/2 in.) plywood. Figure 12 gives the transmission loss for the fan inlet wall, 19.1 mm (3/4 in.) plywood, with the fan inlet hole closed. The attenuation of noise by the air intake duct is above 30 dB for frequencies above 120 Hz as seen in Figure 13. This compares favorably to the design expectation of 25 dB.

For each of the walls, note that the transmission loss exceeds 30 dB for frequencies above 230 Hz.

Directivity

To verify the symmetry of the free-field within the chamber directivity measurements were performed. The test set-up employed is illustrated in Figure 14. The loudspeaker was positioned at the end of a 20 cm (8 in.) diameter fan duct in the center of the fan inlet. The loudspeaker was driven with pure tones at constant voltage ranging from 100 Hz to 1kHz. The microphone was positioned one meter (3.3 ft.) from the center of the loudspeaker on a swinging boom. The boom's position was controlled by the stepping motor shown. The microphone was calibrated with a B & K type 4220 Pistonphone. Measurements were performed at 15-degree increments within the forward 180 degree arc of the loudspeaker.

The results obtained are illustrated in Figures 15 and 16. The range of tones is representative of the range of fundamental blade passage frequencies of possible test rotors. Aside from anomalies due to source directivity and wall interaction effects, the sound field is very uniform at all frequencies including those for which the chamber is only weakly anechoic (i.e. below 150 Hz).

V. CONCLUSIONS

1. Based on the acoustic calibration, the ARL/FED anechoic chamber is moderately anechoic for frequencies above 230 Hz and semi-anechoic for lower frequencies.
2. Inverse square law tests indicate spherical spreading for frequencies down to 200 Hz, indicative of a free field condition. For frequencies down to 100 Hz, the spreading is moderately spherical.
3. The transmission loss provided by the chamber exceeds 30 dB for all frequencies of interest.
4. The chamber may be used with or without the Axial Flow Research Fan and can be configured for other ducts up to a 65 cm (25 in.) diameter.

REFERENCES

1. Lauchle, G. C. and Wong, E., "The ARL/FEU Semi-Anechoic Chamber, " ARL Technical Memorandum, File No. 75-230, 29 September 1975, (Unclassified)
2. Beranek, L. L. and Sleeper, H. P., Jr., "The Design and Construction of Anechoic Sound Chamber," The Journal of the Acoustical Society of America, Vol. 18, No. 1, pp 140-150, July 1976
3. Skudrzyk, Eugene, The Foundations of Acoustics: Basic Mathematics and Basic Acoustics, Springer - Verlag, New York - Wien, 1971, p. 344-375.

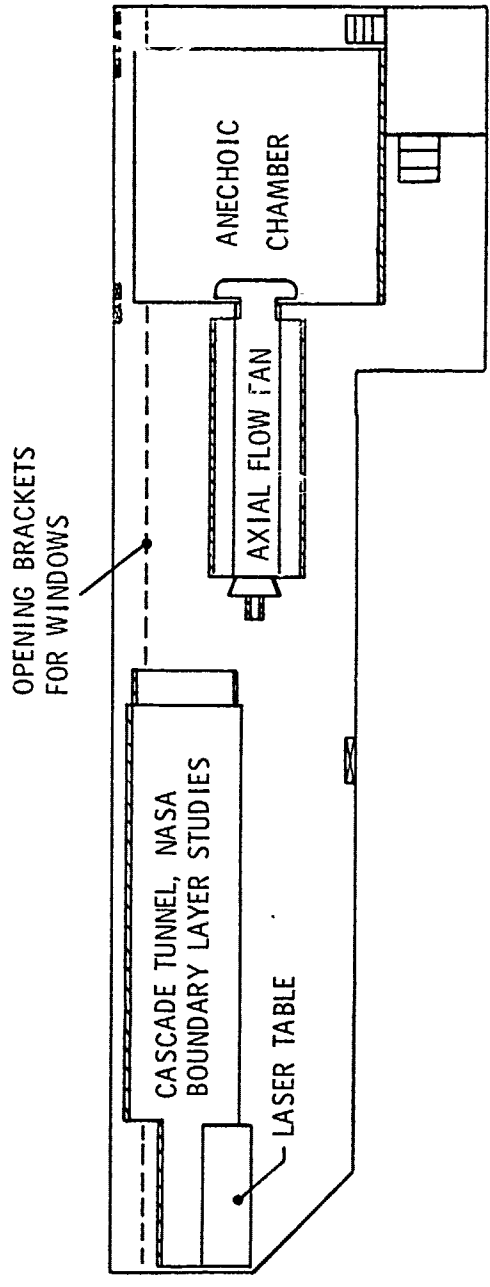


Figure 1 Location of Anechoic Chamber on Above-Console Deck

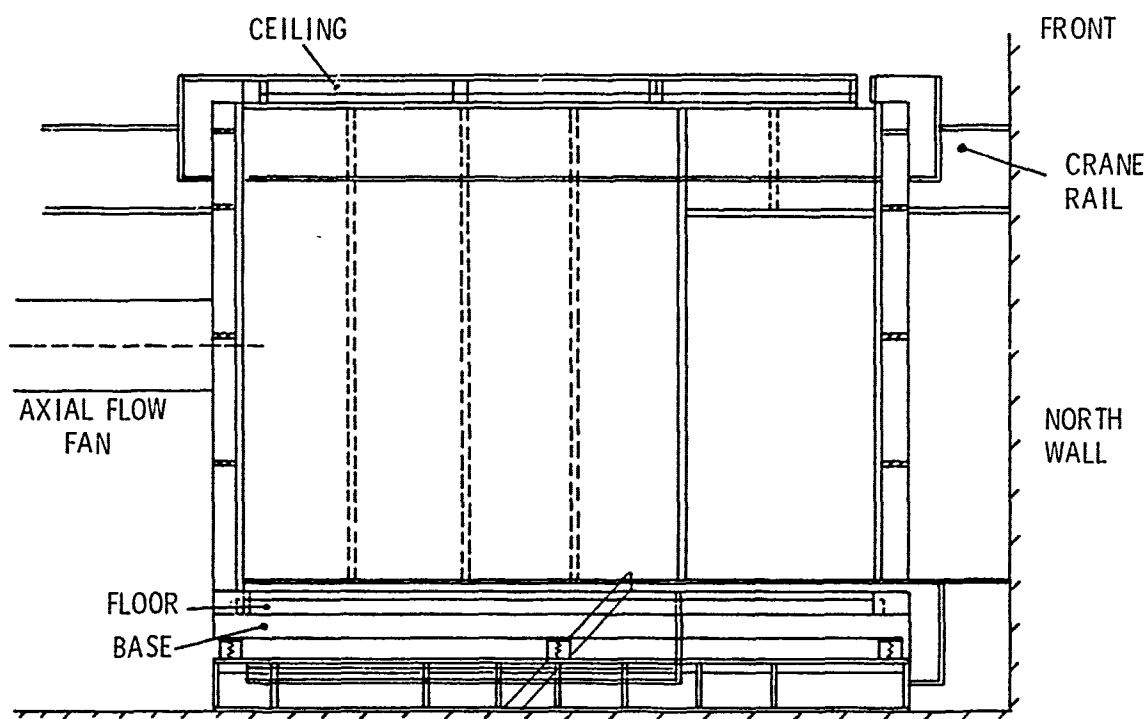


Figure 2 Front View of Anechoic Chamber

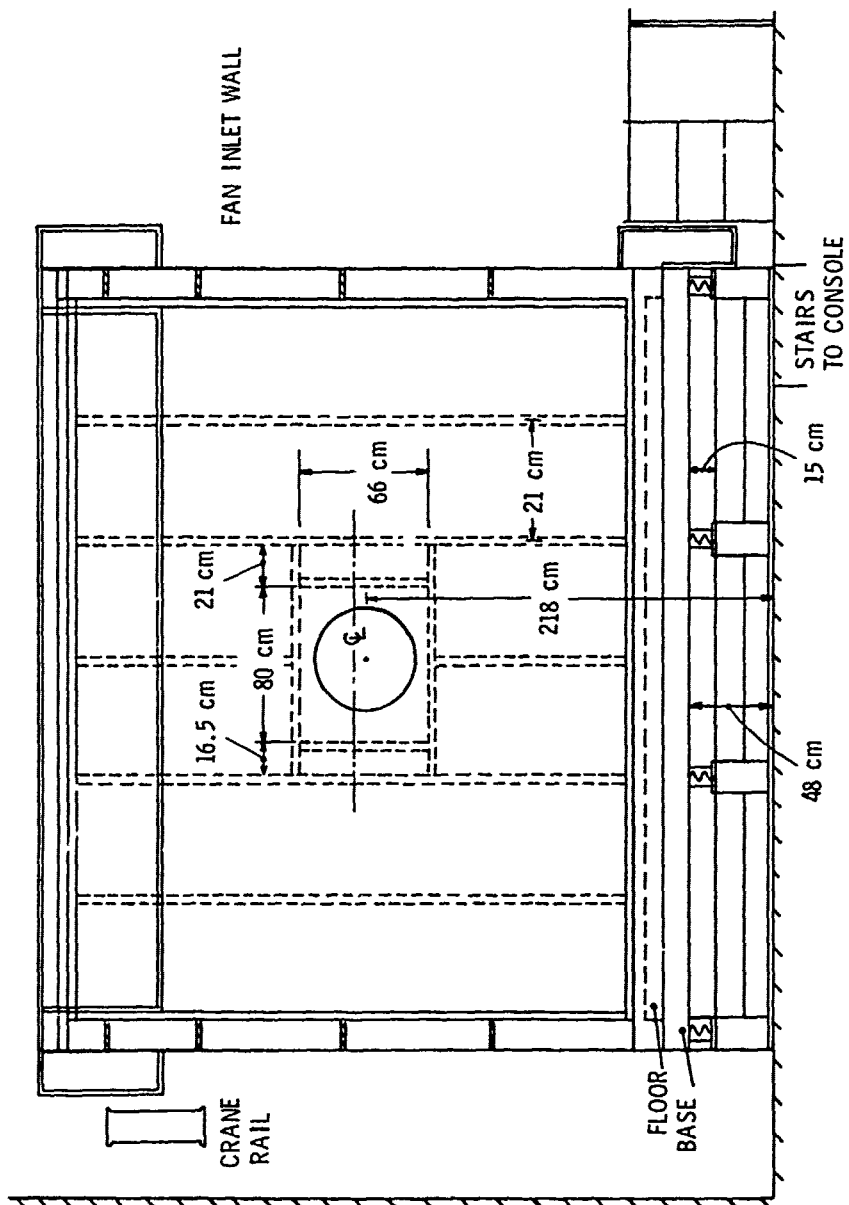


Figure 3 Fan Inlet Wall

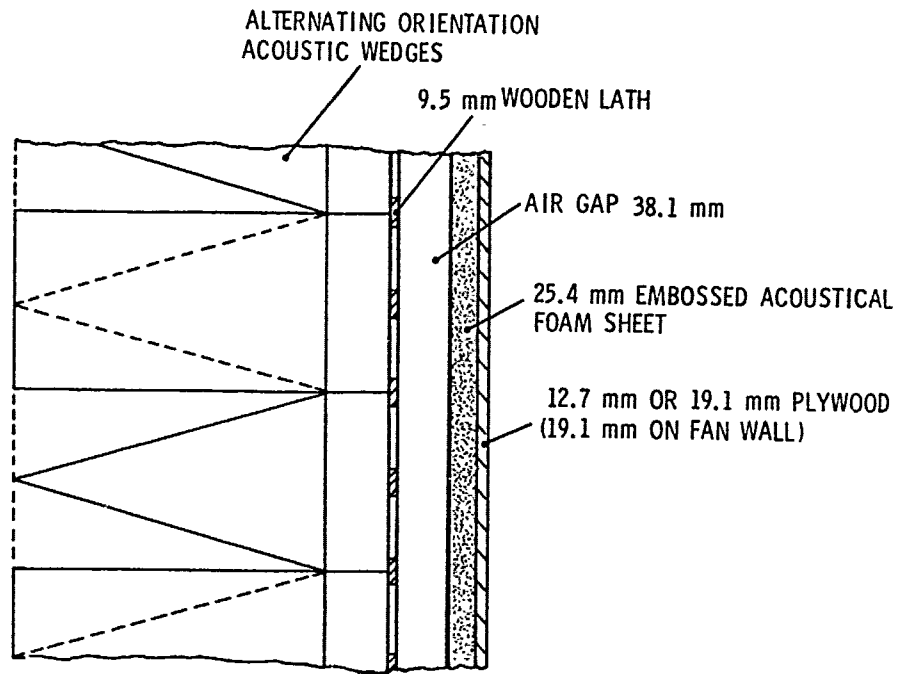


Figure 4 Cross Section of Chamber Wall

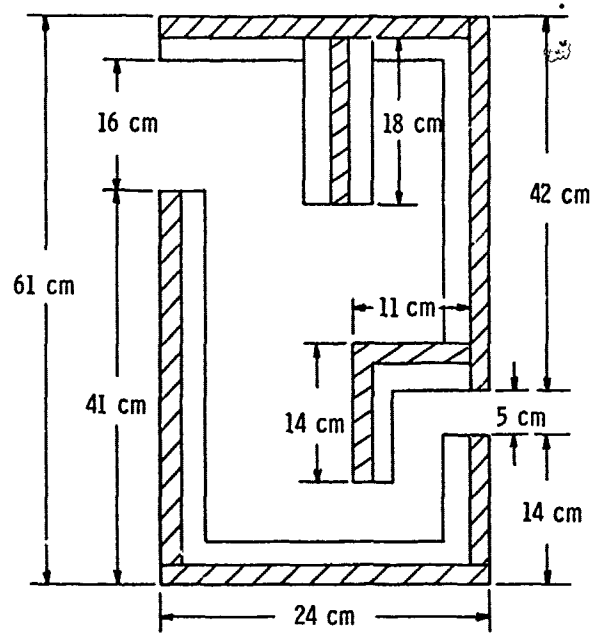


Figure 5 Typical Cross Section of Air Intake Duct

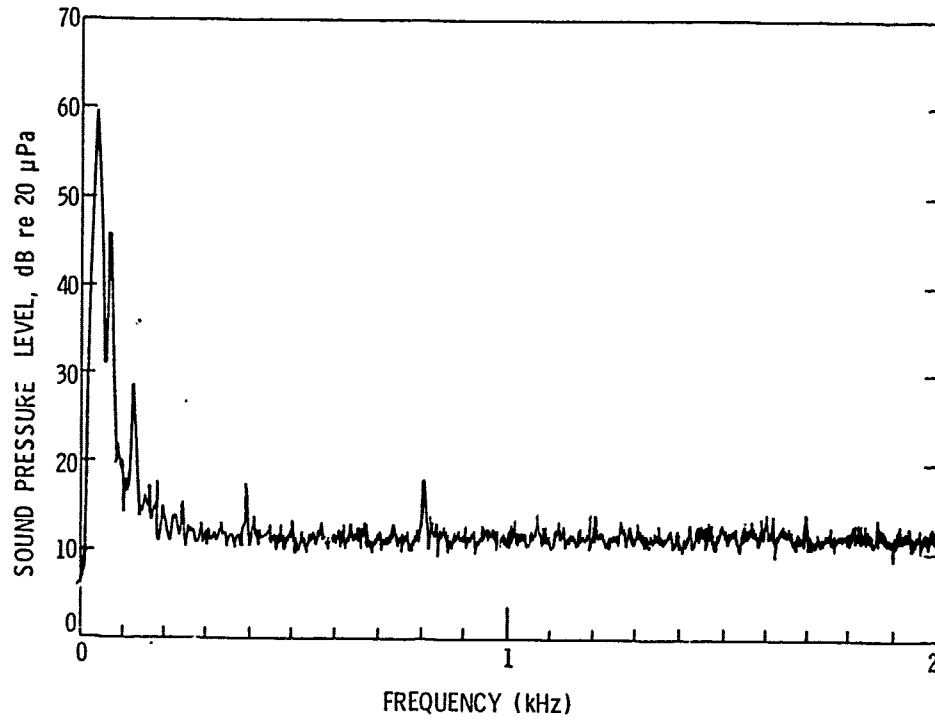
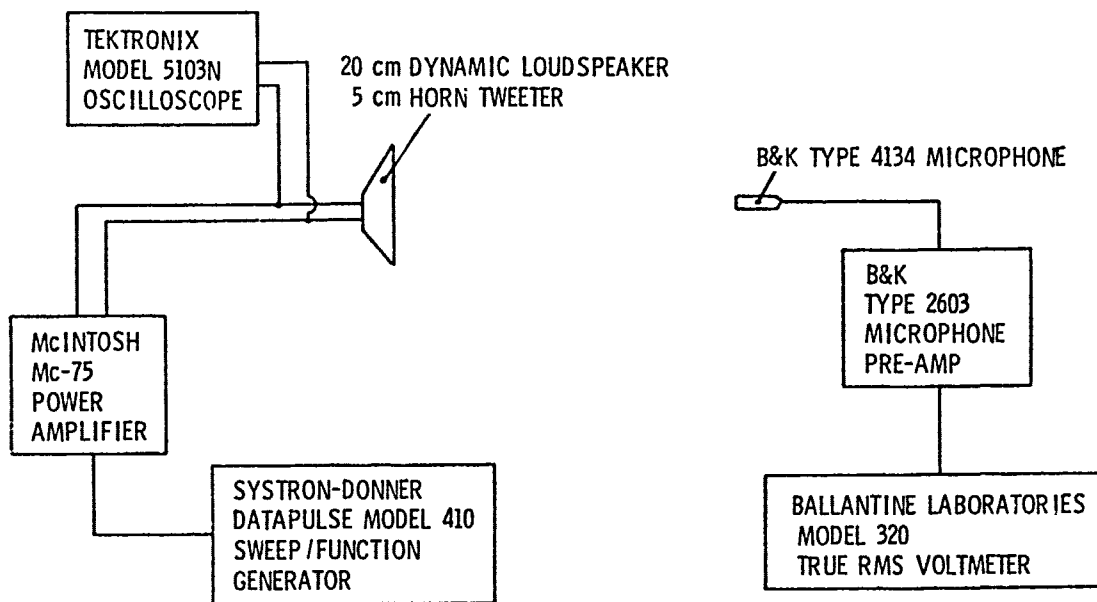


Figure 6 Ambient Noise Sound Pressure Spectrum



NOTE: MICROPHONE CALIBRATION CONDUCTED WITH B&K TYPE 4220 PISTONPHONE

Figure 7 Instrumentation Schematic For Inverse Square Law Measurement

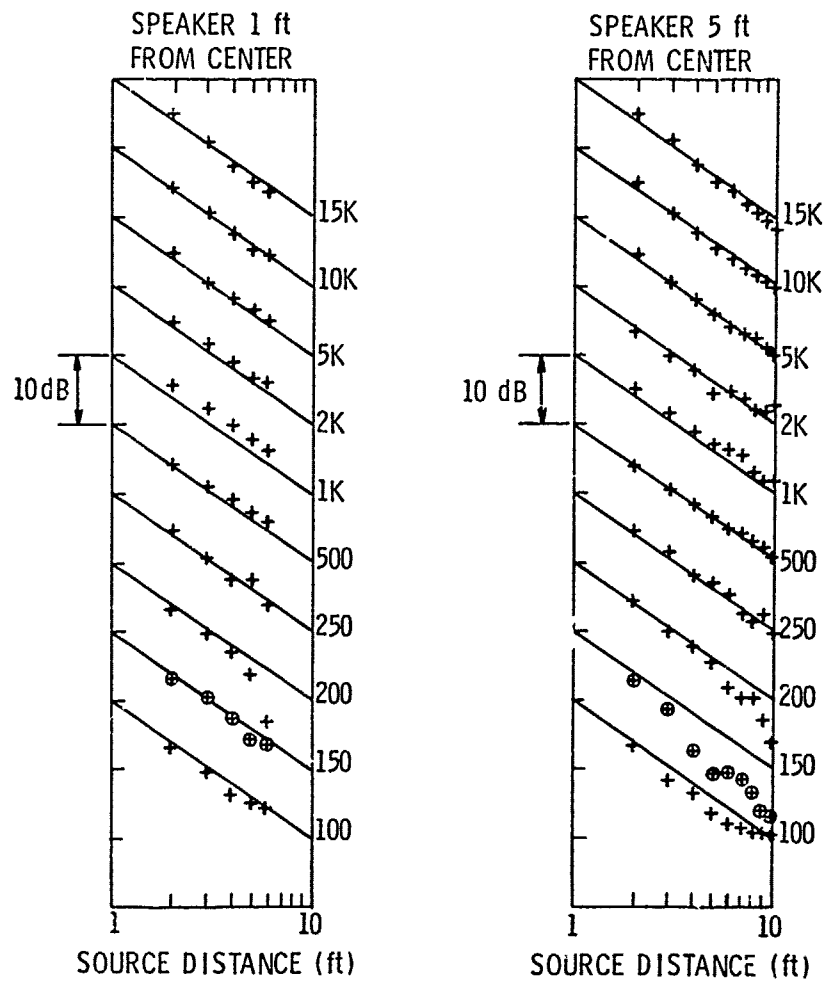


Figure 8 Inverse Square Law Measurement for Selected Tones

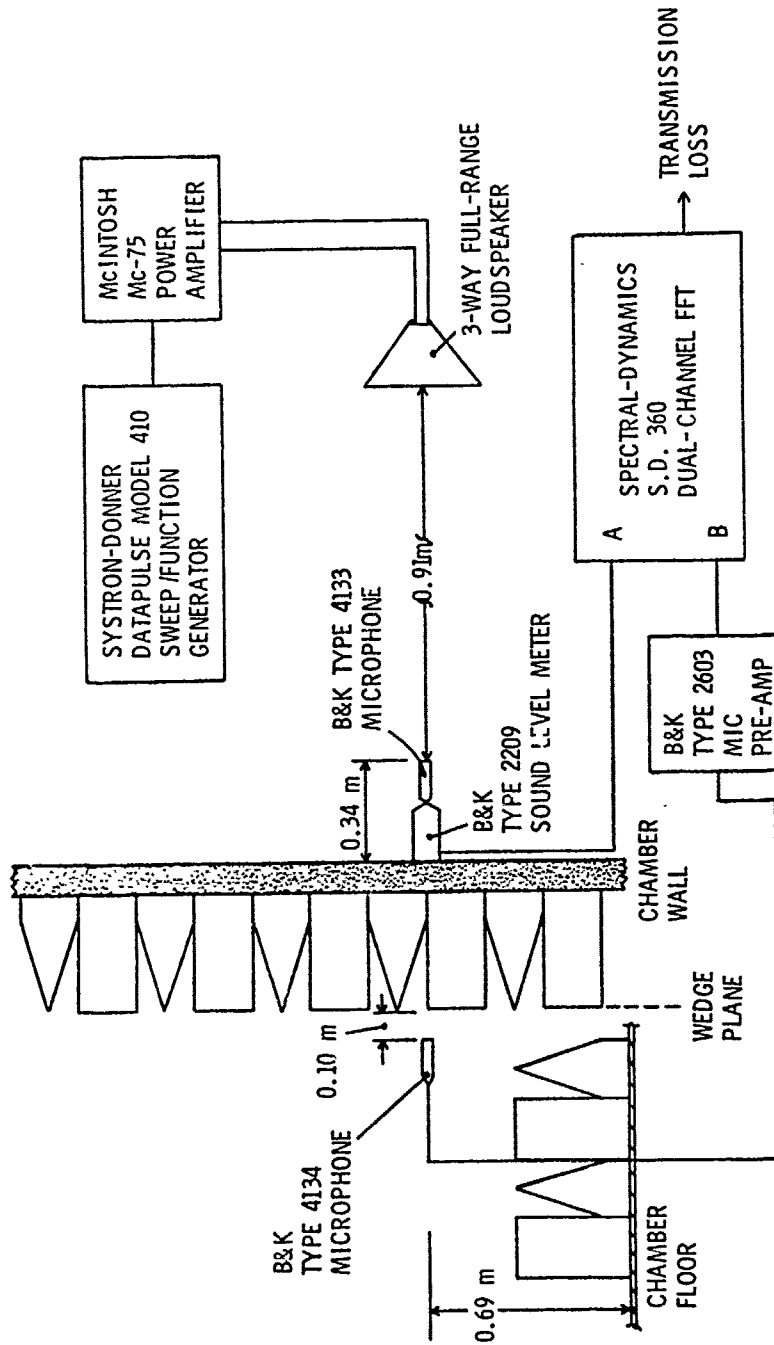


Figure 9 Instrumentation Schematic For Transmission Loss Measurement

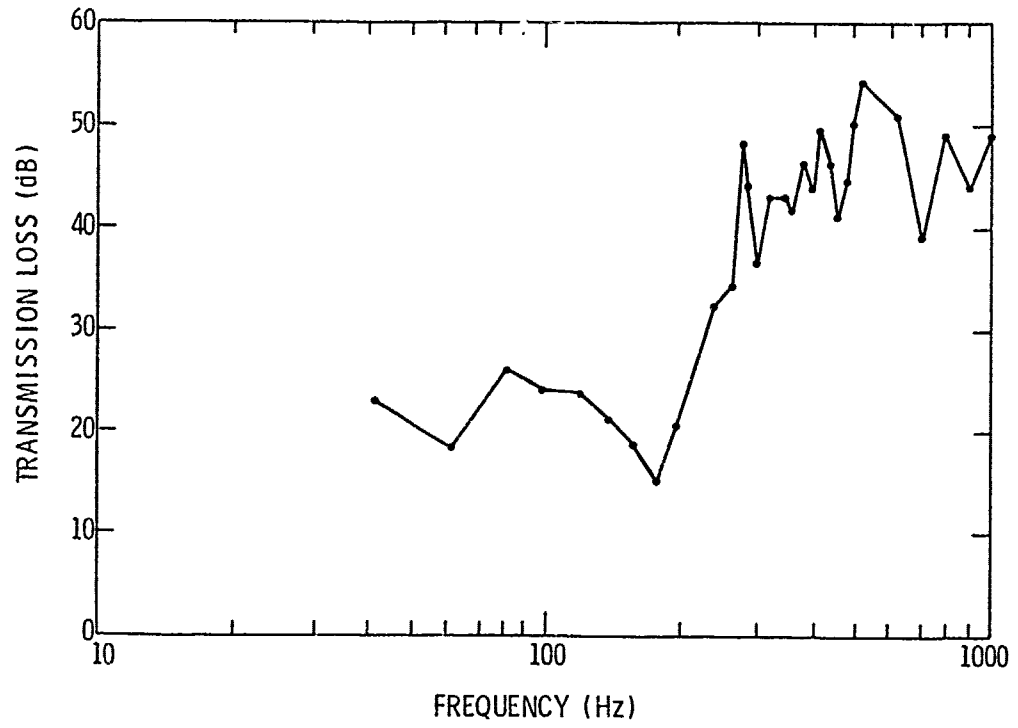


Figure 10 Transmission Loss for Side Wall (Away from Door)

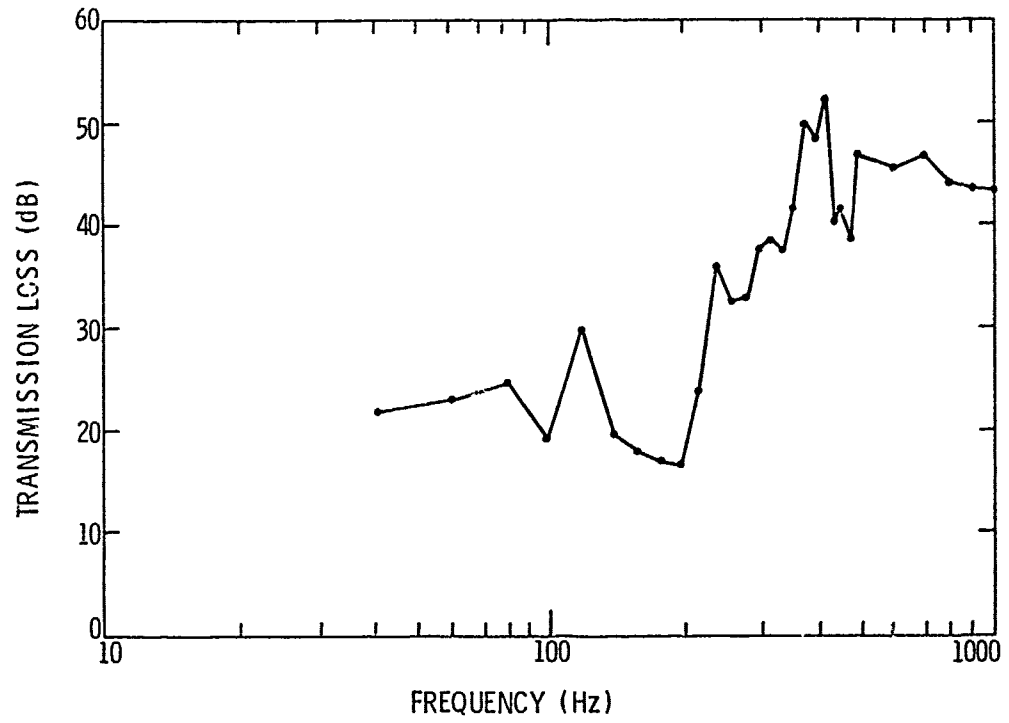


Figure 11 Transmission Loss Through the Door

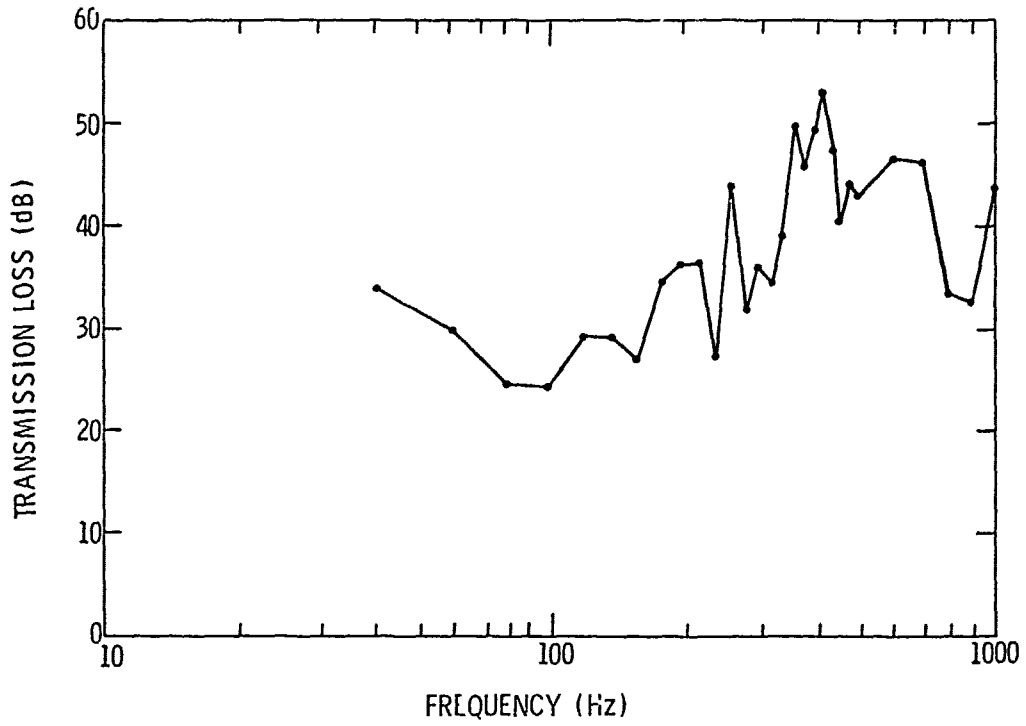


Figure 12 Transmission Loss for Fan Inlet Wall
(Fan duct closed)

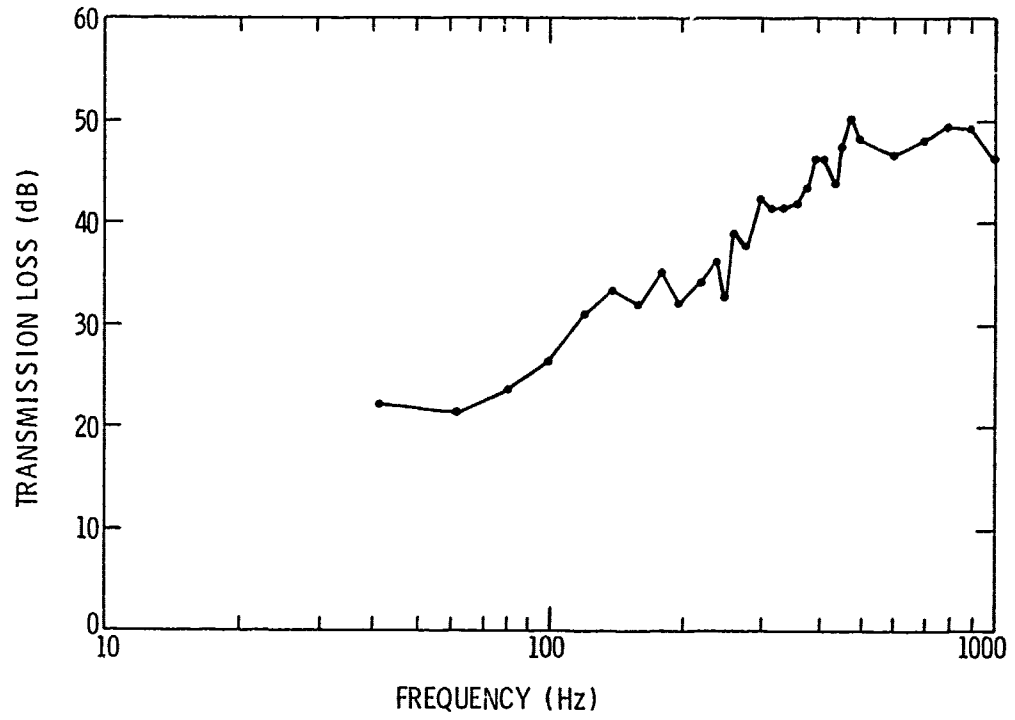


Figure 13 Transmission Loss for Air Intake Duct (3 ft. from end)

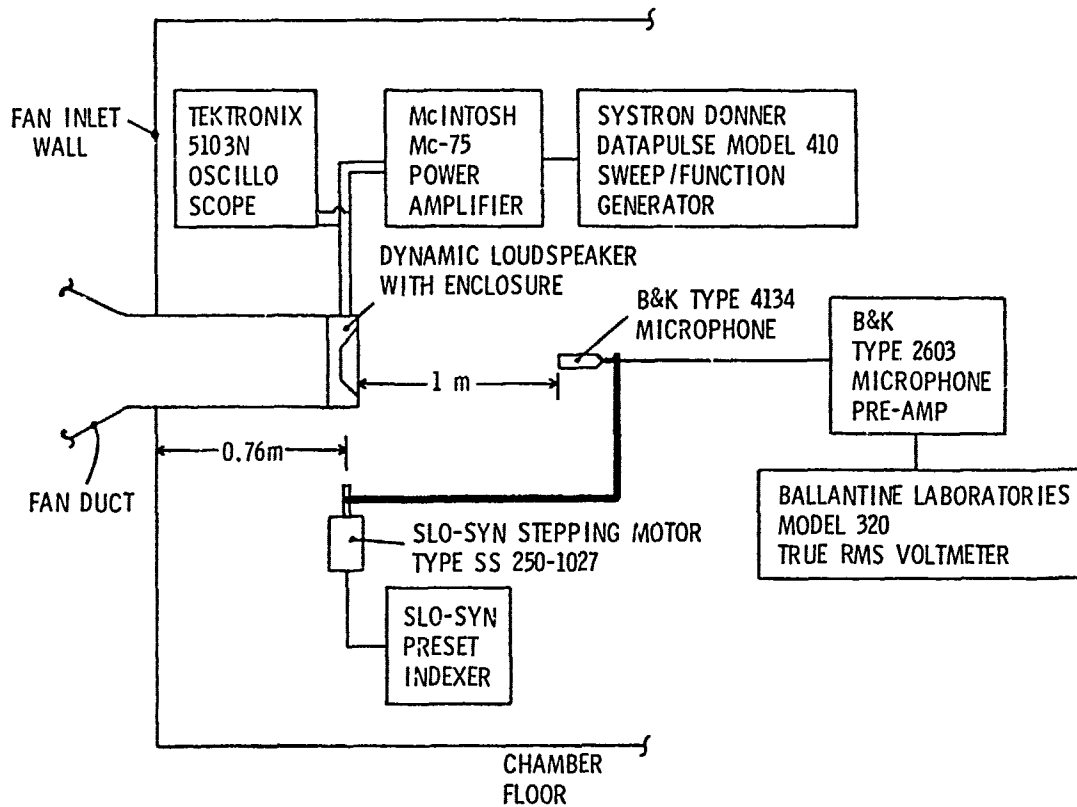


Figure 14 Instrumentation Schematic for Directivity Measurement (Side View)

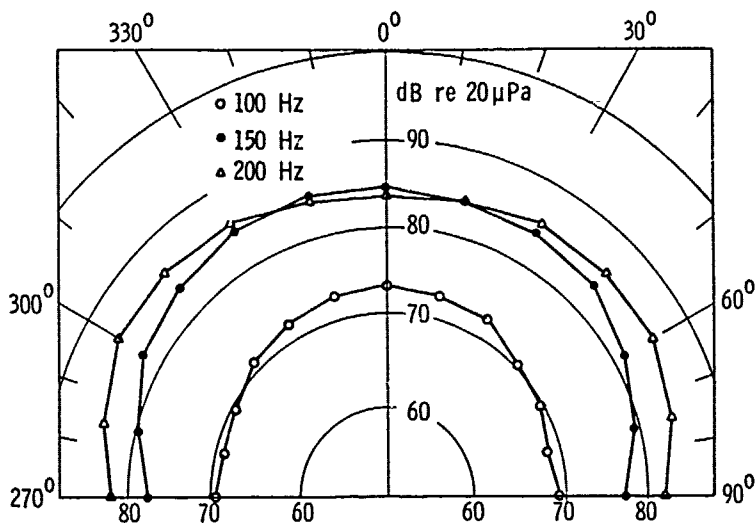


Figure 15 Tonal Directivity Patterns at 100 Hz,
150 Hz and 200 Hz
(source positioned at fan inlet)

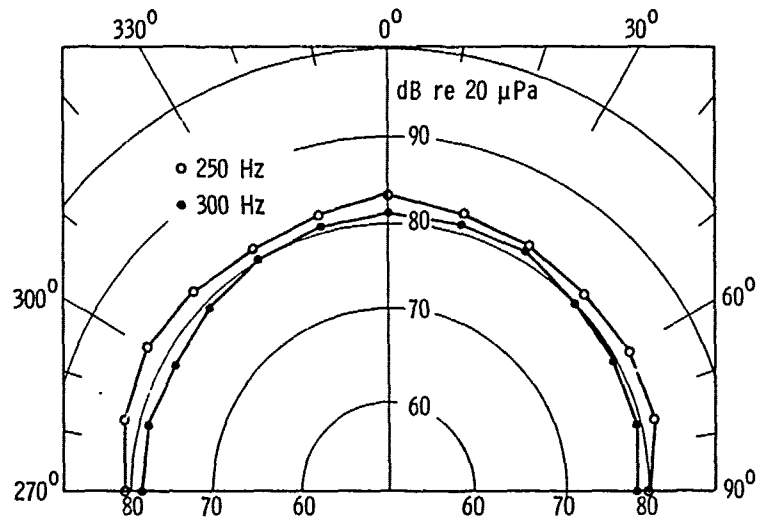


Figure 16 Tonal Directivity Patterns at 250 Hz and 300 Hz (source positioned at fan inlet)

DISTRIBUTION LIST FOR UNLIMITED UNCLASSIFIED TM 81-164 by R. Marboe and J. Fitzgerald
Dated 4 August 1981

Commander
Naval Sea Systems Command
Department of the Navy
Washington, DC 20362
Attn: Library
Code NSEA - 09G32
(Copy Nos. 1 and 2)

Naval Sea Systems Command
Attn: S. M. Blazek
Code NSEA-05HB
(Copy No. 3)

Naval Sea Systems Command
Attn: L. Herstein
Code NSEA-05H
(Copy No. 4)

Naval Sea Systems Command
Attn: F. E. Eissing
Code NSEA-05H
(Copy No. 5)

Naval Sea Systems Command
Attn: E. G. Liszka
Code NSEA-63R1
(Copy No. 6)

Naval Sea Systems Command
Attn: T. E. Peirce
Code NSEA-63R3
(Copy No. 7)

Naval Sea Systems Command
Attn: A. R. Paladino
Code NSEA-05H1
(Copy No. 8)

Commanding Officer
Naval Underwater Systems Center
Newport, RI 02840
Attn: D. Goodrich
Code 3634
(Copy No. 9)

Naval Underwater Systems Center
Attn: R. J. Kittredge
Code 36301
(Copy No. 10)

Naval Underwater Systems Center
Attn: M. Cincotta
Code 36315
(Copy No. 11)

Naval Underwater Systems Center
Attn: C. Hervey
Code 3634
(Copy No. 12)

Naval Underwater Systems Center
Attn: B. J. Meyers
Code 36311
(Copy No. 13)

Naval Underwater Systems Center
Attn: D. A. Quadrini
Code 36314
(Copy No. 14)

Naval Underwater Systems Center
Attn: R. H. Nadolink
Code 3634
(Copy No. 15)

Commanding Officer
Naval Ocean Systems Center
San Diego, CA 92152
Attn: D. Nelson
Code 6342
(Copy No. 16)

Commander
David W. Taylor Naval Ship R & D Center
Department of the Navy
Bethesda, MD 20084
Attn: W. K. Blake
Code 1905
(Copy No. 17)

David W. Taylor Naval Ship R&D Center
Attn: R. W. Brown
Code 1942
(Copy No. 18)

David W. Taylor Naval Ship R & D Center
Attn: T. T. Huang
Code 1552
(Copy No. 19)

DISTRIBUTION LIST FOR UNLIMITED UNCLASSIFIED TM 81-164 by R. Marboe and J. Fitzgerald
Dated 4 August 1981

David W. Taylor Naval Ship R&D Center
Attn: J. H. McCarthy
Code 154
(Copy No. 20)

David W. Taylor Naval Ship R&D Center
Attn: T. C. Mathews
Code 1942
(Copy No. 21)

David W. Taylor Naval Ship R&D Center
Attn: W. B. Morgan
Code 15
(Copy No. 22)

David W. Taylor Naval Ship R&D Center
Attn: K. D. Remmers
Code 1942
(Copy No. 23)

David W. Taylor Naval Ship R&D Center
Attn: T. E. Brockett
Code 1544
(Copy No. 24)

David W. Taylor Naval Ship R&D Center
Attn: F. E. Geib
Code 1942
(Copy No. 25)

David W. Taylor Naval Ship R&D Center
Attn: M. M. Sevik
Code 19
(Copy No. 26)

David W. Taylor Naval Ship R&D Center
Attn: J. Shen
Code 194
(Copy No. 27)

David W. Taylor Naval Ship R&D Center
Attn: F. S. Archibald
Code 1942
(Copy No. 28)

Commander
Naval Surface Weapons Center
Silver Spring, MD 20910
Attn: G. C. Guanaurd
Code R-31
(Copy No. 29)

Office of Naval Research
Department of the Navy
800 North Quincy Street
Arlington, VA 22217
Attn: R. Whitehead
(Copy No. 30)

Office of Naval Research
Attn: H. Fitzpatrick
Code 438
(Copy No. 31)

Office of Naval Research
Attn: A. H. Gilmore
(Copy No. 32)

Office of Naval Research
Attn: M. M. Reishmann
(Copy No. 33)

National Bureau of Standards
Aerodynamics Section
Washington, DC 20234
Attn: P. S. Klebanoff
(Copy No. 34)

National Bureau of Standards
Aerodynamics Section
Washington, DC 20234
Attn: D. S. Pallet
(Copy No. 35)

Officer-in-Charge
David W. Taylor Naval Ship R&D Center
Department of the Navy
Annapolis Laboratory
Annapolis, MD 21402
Attn: E. R. Quandt
Code 272
(Copy No. 36)

David W. Taylor Naval Ship R&D Center
Attn: J. G. Stricker
Code 2721
(Copy No. 37)

David W. Taylor Naval Ship R&D Center
Attn: J. Pierpoint
(Copy No. 38)

DISTRIBUTION LIST FOR UNLIMITED UNCLASSIFIED TM 81-164 by R. Marboe and J. Fitzgerald
Dated 4 August 1981

David W. Taylor Naval Ship R & D Center
Attn: J. Henry
(Copy No. 39)

Naval Research Laboratory
Washington, DC 20390
Attn: R. J. Hansen
(Copy No. 40)

Dynamics Technology, Inc.
22939 Hawthorne Blvd., Suite 200
Torrance, CA 90505
Attn: W. W. Haigh
(Copy No. 41)

Dynamics Technology, Inc.
Attn: G. L. Donohue
(Copy No. 42)

Defense Technical Information Center
5010 Duke Street
Cameron Station
Alexandria, VA 22314
(Copy No. 43)

Dr. A. J. Acosta
Hydrodynamics Lab.
California Institute of Technology
Pasadena, CA 91100
(Copy No. 44)

Dr. C. Ball
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 46)

Dr. W. Lang
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 47)

Dr. G. Maling, Jr.
IBM Research Laboratories
Poughkeepski, N.Y.
(Copy No. 48)

Dr. Tom Booth
93-355-503-4A
Garrett Turbine Engine Company
111 South 34th Street
P.O. Box 5217
Phoenix, AZ 85010
(Copy No. 49)

Dr. James H. Dittmar
MS 501-4
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 50)

Dr. R. P. Dring
United Technologies Research Lab.
400 Main Street
East Hartford, CT 06108
(Copy No. 51)

Mr. C. Feiler
MS 501-4
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 52)

Mr. Philip R. Gliebe
General Electric Company
Aircraft Engine Group
Cincinnati, OH 45215
(Copy No. 53)

Dr. M. E. Goldstein
NASA Lewis Research Center
MS 301-1
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 54)

Dr. John Gorenweg
Acoustics Division
NASA Lewis Research Center
21000 Brook Park Road
Cleveland, OH 44135
(Copy No. 55)

Dr. Don Hanson
Mail Stop 1A-3-6
Hamilton Standard
Windsor Locks, CT 06096
(Copy No. 56)

Mr. M. J. Hartmann
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 57)

DISTRIBUTION LIST FOR UNLIMITED UNCLASSIFIED TM 81-164 by R. Marboe and J. Fitzgerald
Dated 4 August 1981

Dr. Loretta M. Shaw
MS 501-4
Nasa Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 58)

Dr. W. H. Heiser
General Electric Company
Aircraft Engine Division
Cincinnati, OH 45215
(Copy No. 59)

Dr. G. Homicz
Calspan Corporation
Buffalo, NY 14221
(Copy No. 60)

Mr. J. D. Kester
Engineering Building
Pratt & Whitney Aircraft
East Hartford, CT 06108
(Copy No. 61)

Dr. William McNally
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 62)

Dr. A. A. Mikolaczak
Pratt & Whitney Aircraft
400 Main Street
East Hartford, CT 06118
(Copy No. 63)

Dr. T. H. Okiishi
Dept. of Mechanical Engineering
Iowa State University
Ames, IA 50010
(Copy No. 64)

Dr. Peter Runstadler, Jr.
Creare Inc.
Box 71
Hanover, NH 03755
(Copy No. 65)

Dr. L. H. Smith
Turbomachinery Development Unit
Applied Res. Operations Bldg. 300
General Electric Company
Cincinnati, OH 45215
(Copy No. 66)

Mr. P. M. Sociol
Fluid Components Division
NASA Lewis Research Center
21000 Brookpark Road
Cleveland, OH 44135
(Copy No. 67)

Mr. J. R. Switzer
Garrett Turbine Engine Company
111 South 34 Street
P.O. Box 5217
Phoenix, AZ 85010
(Copy No. 68)

Dr. A. J. Wennerstrom
AFWAL/POTX
Aero Propulsion Lab.
Wright Patterson Air Force Base,
OH 45433
(Copy No. 69)

Applied Research Laboratory
The Pennsylvania State University
Post Office Box 30
State College, PA 16801
Attn: R. E. Henderson
(Copy No. 70)

Applied Research Laboratory
Attn: D. E. Thompson
(Copy No. 71)

Applied Research Laboratory
Attn: B. R. Parkin
(Copy No. 72)

Applied Research Laboratory
Attn: R. C. Marboe
(Copy No. 73)

Applied Research Laboratory
Attn: J. M. Fitzgerald
(Copy No. 74)

Applied Research Laboratory
Attn: G. C. Lauchle
(Copy Nos. 75 & 76)

Applied Research Laboratory
Attn: B. Lakshminarayana
(Copy Nos. 77 & 78)

DISTRIBUTION LIST FOR UNLIMITED UNCLASSIFIED TM 81-164 by R. Marboe and J. Fitzgerald
Dated 4 August 1981

Applied Research Laboratory
Attn: J. Tichy
(Copy No. 79)

Applied Research Laboratory
Attn: W. Thompson, Jr.
(Copy No. 80)

Applied Research Laboratory
Attn: J. Maynard
(Copy No. 81)

Applied Research Laboratory
Attn: M. T. Pigott
(Copy No. 82)