

AD-A094 984

NAVAL POSTGRADUATE SCHOOL MONTEREY CA
PROPAGATION OF SOUND IN A FAST BOTTOM UNDERLYING A WEDGE-SHAPED-ETC(U)
SEP 80 N A BRADSHAW

F/O 28/1

UNCLASSIFIED

NL

1 of 1

AD-A094 984



END
DATE
FILMED
DTIC

AD A094984

NAVAL POSTGRADUATE SCHOOL
Monterey, California



THESIS

PROPAGATION OF SOUND IN A FAST BOTTOM UNDERLYING
A WEDGE-SHAPED MEDIUM

by

Norine A. Bradshaw

September 1980

Thesis Advisor:

A. B. Coppens

Approved for public release; distribution unlimited

UDC FILE COPY

81 2 13 047

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER	2. GOVT ACCESSION NO. AD-A094984	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) Propagation of Sound in a Fast Bottom Underlying a Wedge-Shaped Medium .	5. TYPE OF REPORT & PERIOD COVERED Master's Thesis September 1980	6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(s) Norine A. Bradshaw	8. CONTRACT OR GRANT NUMBER(s)	
9. PERFORMING ORGANIZATION NAME AND ADDRESS Naval Postgraduate School Monterey, California 93940	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS	
11. CONTROLLING OFFICE NAME AND ADDRESS Naval Postgraduate School Monterey, California 93940	12. REPORT DATE September 1980	13. NUMBER OF PAGES 82
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office) Naval Postgraduate School Monterey, California 93940	15. SECURITY CLASS. (of this report) Unclassified	16a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) acoustics sound propagation continental shelf method of images		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The pressure and phase distribution of sound in a fast fluid medium underlying a tapered fluid medium was modeled using a Green's function approach. The model predicted well-defined beams in the bottom, with apparent interference effects strongly evident for cases at a distance from the wedge interface. The effects of attenuation on the patterns were studied. A simple expression for determining the beam depression angle as a function, the wedge angle, the sound speed ratio, and the density ratio was derived. Comparisons with laboratory measurements were made.		

201450

Approved for public release; distribution unlimited

Propagation of Sound in a Fast Bottom Underlying
a Wedge-Shaped Medium

by

Norine A. Bradshaw
Lieutenant Commander, United States Naval Reserve
B.S., Michigan State University, 1968

Submitted in partial fulfillment of the
requirements for the degree of

MASTER OF SCIENCE IN ENGINEERING ACOUSTICS

from the

NAVAL POSTGRADUATE SCHOOL
September 1980

A

Author:

Norine A. Bradshaw

Approved by:

Alan B. Cooper

Thesis Advisor

James V. Sanders

Second Reader

William B. Zelinger

Chairman, Department of Physics and Chemistry

William M. Jolley

Dean of Science and Engineering

ABSTRACT

The pressure and phase distribution of sound in a fast fluid medium underlying a tapered fluid medium was modeled using a Green's function approach. The model predicted well defined beams in the bottom, with apparent interference effects strongly evident for cases at a distance from the wedge interface. The effects of attenuation on the patterns were studied. A simple expression for determining the beam depression angle as a function the wedge angle, the sound speed ratio, and the density ratio was derived. Comparisons with laboratory measurements were made.

TABLE OF CONTENTS

I. INTRODUCTION -----	6
II. THEORY -----	9
III. DEVELOPMENT OF WDGBTM -----	13
A. EQUATION PREPARATION -----	14
1. Depth Domain -----	14
2. Angle Domain -----	14
3. Design Restrictions -----	15
4. Program Results -----	15
B. SENSITIVITY TO INTEGRATION STEP SIZE -----	16
IV. BEAM DEPRESSION ANGLE ANALYSIS -----	19
V. THE EFFECT OF ATTENUATION -----	22
VI. ANALYSIS OF FAR-FIELD INTERFERENCE PATTERNS -----	31
A. DIFFERENTIAL CHANGE APPROACH -----	31
B. MULTIPLE LAMBDA APPROACH -----	34
C. SUMMARY OF ANALYSES -----	34
VII. COMPARISONS -----	37
A. EXPERIMENTAL DATA -----	37
B. PARABOLIC EQUATION APPROACH -----	41
VIII. CONCLUSIONS -----	43
APPENDIX A - SAMPLE INPUT SET -----	44
APPENDIX B - GREENS FUNCTION RESULTS -----	46
APPENDIX C - BEAM DEPRESSION ANGLE CALCULATIONS -----	51
APPENDIX D - EXPERIMENT/WDGBTM COMPARISONS -----	57

<i>PROGRAM WDGBTM</i> -----	61
<i>LIST OF REFERENCES</i> -----	31
<i>INITIAL DISTRIBUTION LIST</i> -----	32

I. INTRODUCTION

The problem of radiation behavior in a wedge-shaped medium overlying a bottom possessing a greater speed of sound has been investigated both optically and acoustically. In 1971, Tien and Martin (Ref. 1) deposited a thin, tapered, dielectric film on a substrate having a higher refractive index, and observed the behavior of a laser beam coupled into the film. Perfect reflection at the film-substrate interface was observed until the changing angle of incidence decreased below the critical angle for continued mode propagation. For angles of incidence below critical, the light was converted into radiation modes in the substrate.

A concurrent analysis of the acoustic problem was done by Kuznetsov (Ref. 2). Kuznetsov developed a theory, based on normal-modes, that predicted the following for sound propagating within a wedge-shaped medium overlying a half-space with a higher sound speed:

1. As the sound approached the vertex of the wedge, it would continue to be totally reflected until the angle of incidence decreased to the limiting angle of total reflection.

2. Sound incident at less than the limiting angle of total reflection would be totally refracted into the underlying half-space. This refraction occurs within the region from the wedge apex to the point where the limiting angle was attained.

3. Within the half-space, the acoustic energy would be propagated as a well defined beam with the maximum pressure amplitude occurring at an angle of depression from the wedge interface within the range β to 2β , where β is the wedge angle.

Kuznetsov also performed a series of experiments that supported his theory.

Subsequent optical experiments by Tien, Smolinsky, and Martin (Ref. 3) demonstrated the formation of a well-defined beam within the substrate. The accompanying theoretical discussion predicted conflicting behavior. Ray-optics predicted refraction into the substrate after the cutoff distance (the point where refraction begins) and wave theory predicted complete refraction into the substrate before the cutoff point. Sigelman, et. al. (Ref. 4) both predicted and observed a well columnated beam in a water-aluminum wedge system, but the observed beam was much broader than predicted. Also, the pressure maxima for the columnated beam occurred at approximately 11° for a wedge angle of 1.3° , well outside the limit predicted by Kuznetsov.

The primary objective of this theoretical development is to produce a simple, fast computer model to predict the pressure and phase distribution of the columnated sound beam in the fast substrate. The pressure and phase distribution along the wedge interface calculated by Kawamura (Ref. 5) will be used as input to the substrate model, and will specifically be used as the boundary conditions for a Green's function analysis. To remove explicit frequency dependence, all relevant distances will be normalized to the distance measured from the apex along the wedge interface at which the critical reflection angle is first exceeded; this dump

distance is referred to as the dump distance X . Integration over the wedge interface will be truncated at distances sufficient to encompass only selected modes of interest. Results will be analyzed for trends and compared with existing measured data.

II. THEORY

Pressure and phase distributions along the bottom of a wedge-shaped medium are obtained using an adaptation of the method of images as prescribed in Ref. 5. The source may be set at either finite or infinite distance, but must be maintained sufficiently far from the wedge boundaries to ensure that impinging waves are essentially locally planar. The following theoretical discussion uses the above pressure and phase distribution as initial conditions, and calculates the pressure and phase distribution in the fast bottom by utilizing the Green's function (Ref. 6).

Figure II-1 illustrates the geometry of the problem. The horizontal x-axis and the vertical z-axis have their point of origin at the wedge apex, with the positive x-axis forming the interface between the wedge and fast bottom. The wedge angle is designated β and the angle specifying the sound source location is θ . X is the first mode dumping distance, and is a distance determined as follows:

$$\theta_c = \arccos (C_2 / C_1) \quad (\text{II-1a})$$

$$D_0 = C_1 / (4 \cdot f \cdot \sin(\theta_c)) \quad (\text{II-1b})$$

$$X = D_0 / \sin(\beta) \quad (\text{II-1c})$$

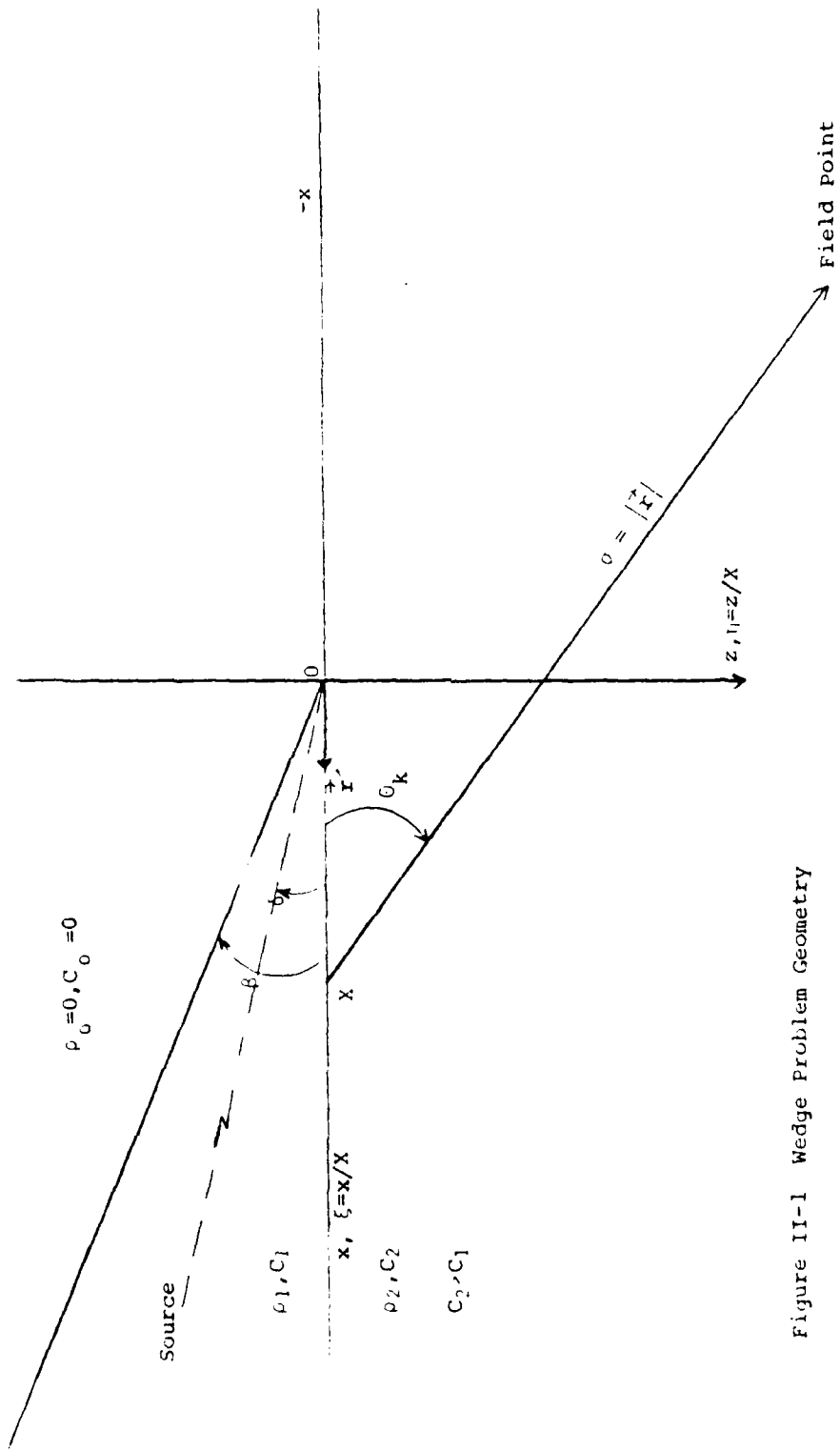


Figure II-1 Wedge Problem Geometry

where

C_1 = sound speed in the wedge

C_2 = sound speed in the bottom

f = frequency

and D_0 = lowest mode cut-off depth.

To make the problem frequency independent, all distances were normalized to X by defining

$$\xi = x / X \quad (\text{II-2a})$$

$$\xi' = x' / X \quad (\text{II-2b})$$

and

$$\eta = z / X \quad (\text{II-2c})$$

The variable x' is identical to x , and ξ' is identical to ξ . The prime designation is required for dummy variables of integration in the Green's function development to follow.

Pressure along the wedge bottom is then given by

$$p(\xi, \eta=0, t) = P(\xi) \exp \{i[\omega t + L(\xi)]\} \quad (\text{II-3})$$

where $P(\xi)$ is the pressure amplitude along the wedge bottom and $L(\xi)$ is the phase along the wedge bottom. Pressure in the bottom, away from the boundary, is then given by Ref. 7,

$$\begin{aligned} p(\vec{r}, t) &= \exp[i\omega t] \frac{1}{4\pi} \int_0^\infty P(\xi') \left. \frac{\partial G}{\partial z'} \right|_{z'=0} x d\xi' \quad (\text{II-4}) \\ &\equiv \exp [i\omega t] R(\vec{r}) \end{aligned}$$

where G is the appropriate Green's function. The z' coordinate is identical to the z coordinate and $\vec{r} = (\vec{x}^2 + \vec{z}^2)^{1/2}$. The appropriate Green's function, for a line source in cylindrical geometry is

$$G(\vec{r}, \vec{r}') = i\pi H_0^{(2)}(k_2 |\vec{r} - \vec{r}'|) \quad (\text{II-5})$$

where k_2 is the propagation vector in the bottom and $\vec{r} = [(\vec{x}')^2 + (\vec{z}')^2]^{1/2}$.

Differentiation of the Green's function yields (Ref. 7)

$$\frac{\partial G}{\partial z'} = i\pi H_1^{(2)}(k_2 |\vec{r} - \vec{r}'|) \frac{k_2 (z - z')}{|\vec{r} - \vec{r}'|} \quad (\text{II-6})$$

where $k_2 |\vec{r} - \vec{r}'| \gg 2\pi$ and $H_0^{(2)}$, $H_1^{(2)}$ are zero and first order Bessel functions of the third kind (Hankel functions) respectively. Using equations (II-2a,b,c) at $z' = 0$ the expression $|\vec{r} - \vec{r}'|$ may be rewritten as

$$|\vec{r} - \vec{r}'| = x [(\xi - \xi')^2 + n^2]^{1/2} \quad (\text{II-7})$$

Substituting (II-7) into equation (II-6), and evaluating $\frac{\partial G}{\partial z'}$ at $z' = 0$ yields

$$\left. \frac{\partial G}{\partial z'} \right|_{z'=0} = \frac{\pi \left(\frac{2}{\pi k_2}\right)^{1/2} \exp\left(\frac{i\pi}{4}\right) \exp\{-ik_2 x [(\xi - \xi')^2 + n^2]^{1/2}\} k_2 n x}{x^{3/2} [(\xi - \xi')^2 + n^2]^{3/4}} \quad (\text{II-8})$$

Combining equations (II-8) and (II-4) yields a time independent expression for amplitude and phase, $R(\vec{r})$

$$R(\vec{r}) = \frac{n}{4} \exp\left(\frac{i\pi}{4}\right) \left(\frac{2k_2 x}{\pi}\right)^{1/2} \int_0^\infty \frac{P(\xi') \exp\{i[L(\xi') - k_2 x [(\xi - \xi')^2 + n^2]^{1/2}]\} d\xi'}{[(\xi - \xi')^2 + n^2]^{3/4}} \quad (\text{II-9})$$

Equation (II-9) is the basic equation used for the calculation of phase and amplitude in the bottom.

III. DEVELOPMENT OF WDGBTM

Program WDGBTM was designed to perform four main functions. First, the pressure and phase along the interface of a wedge and underlying fast media may be calculated in two ways, for both infinitely and finitely distant sound sources in accordance with Ref. 5. Modifications were made to the code of Ref. 5 to facilitate subsequent in-bottom calculations, and to enhance user flexibility. These changes did not affect results. Then using the first code as input, additional code enables both the computation of pressure and phase along a line perpendicular to the wedge bottom, and as a function of radius and angle measured from the first mode dumping point, X.

Modes of program execution may be chosen by literal input directives combined with appropriate numerical input. Appendix A illustrates an input set that utilizes all four options. Output consists of both printed listings and graphics.

Program versions exist for execution on both a CDC 6500 and an IBM 360. The CDC 6500, with a 60 bit word, does not require double precision for accuracy equivalent to the IBM 360 output. The CDC version of WDGBTM has simpler varian control routines, and separate output files for each pressure/phase variation. This enables optional *suppression* of listings. Execution time on the CDC 6500 is roughly ten times faster than on the IBM 360. For average input sets, times are 150 and 1500 cp seconds respectively.

A. EQUATION PREPARATION

1. Depth Domain

Integrations along lines perpendicular to the wedge bottom were designated depth domain calculations. The program was designed to allow specification of the field point, i.e., specification of where along the media interface line the perpendicular probe line could be dropped. The field point is not confined to lines directly beneath the wedge, but may extend to points beyond the wedge apex. These points would be values of negative x in Figure II-1. The beginning and ending points of the section of the perpendicular to be examined, and the number of field points used in the interval were also to be specified.

Equation (II-9) was split into real and imaginary components before coding. Integration order selected was over the entire ξ' domain for each η point.

2. Angle Domain

Examination of Figure II-1 verifies the following geometric relationships.

$$\varepsilon = \sigma \cos\theta_k \quad (\text{III-1a})$$

and

$$\eta = (\sigma^2 - \varepsilon^2)^{1/2} \quad (\text{III-1b})$$

where ε is the distance from X to the ξ' coordinate of the field point of interest and σ is the radial distance from X to the field point.

Substitution of equations (III-1a) and (III-1b) into equation (II-9)

yields

$$R(\vec{r}) = \frac{(c^2 - \epsilon^2)^{1/2}}{4} \exp\left(\frac{i\pi}{4}\right) \left(\frac{2k_2 X}{\pi}\right)^{1/2} \int_0^{\vec{r}} \frac{\vec{P}(\xi') \exp\{i[L(\xi') - k_2 X [(\xi - \xi')^2 + c^2 - \epsilon^2]^{1/2}]\} d\xi'}{[(\xi - \xi')^2 + c^2 - \epsilon^2]^{3/4}} \quad (\text{III-2})$$

Equation (III-2) is the basis for angle domain calculations. The specification of \vec{r} originating at X was chosen to conform to existing experimental data (Reference 8). The length of the radius, the beginning and ending of the angular sweep, and the angular increment comprise the input specifications. As in the depth domain option, equation (III-2) was split into real and imaginary components before coding.

3. Design Restrictions

Theoretical development of program WDGBTM assumes the sound source is sufficiently distant to ensure essentially locally planar waves at the interface and the in-bottom measurements satisfy the condition $k_2 |\vec{r} - \vec{r}'| \gg 2\pi$.

4. Program Results

An initial set of WDGBTM runs were made in the depth domain for a perpendicular directly down from the wedge apex. The source was infinitely distant and set at an angle of $\beta/2$. This limited modes within the wedge to odd numbered (symmetric) modes only. Integration over the wedge bottom was truncated at ξ equal four, so that only modes one and three were considered as contributing significantly to the beam pattern in the bottom. *Beam patterns in the bottom were calculated over the range η equal zero to 1.3X.*

The initial set was comprised of seventy-five runs. In addition to the above criteria, the runs were generated by all possible combinations of the following specifications. The ration ρ_1/ρ_2 was 0.95, 0.90, 0.80, 0.70 or 0.50. The sound speed ratio, C_1/C_2 was set at 0.95, 0.90, or 0.85. The wedge angle β was varied through 10, 6.92, 5, 2 and 1 degree(s).

From the results of the initial data set, the depression angle and beam width was calculated for the beam pattern in the bottom. Results are summarized in Appendix B. These results were then used to find a simple analytical formula to predict the depression angle as discussed in Section IV.

Other runs were made for perpendiculars at $x = -9X$ from the wedge apex. These runs displayed apparent strong interference phenomena and are discussed in detail later.

Angle domain executions were made to compare with experimental data. Results are discussed in Section VII.

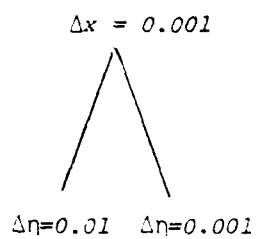
B. SENSITIVITY TO INTEGRATION STEP SIZE

It was desirable to keep both the integration step size and the field point increment size as large as possible in order to minimize total computer resources required. Increment of 0.01 times the dump distance (X) proved to be optimum for both integration increments along the wedge interface and field point steps in the bottom.

To illustrate the adequacy of 0.01X increment, a series of runs were made in accordance with Table III-1. An overlay of wedge interface pressure amplitude for $\Delta x = 0.01X$ and $\Delta x = 0.001X$ demonstrated no distinguishable difference. Likewise, depth domain plots (a, b, c, d of

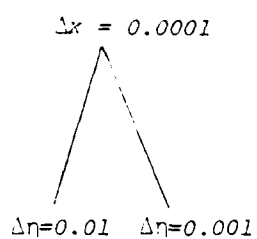
Table (III - 1)

Sensitivity to Integration Step Size (Δx) and Field point interval size ($\Delta \eta$)



(a)

(b)



(c)

(d)

Run	η amp max	θ_D
a	0.240	13.50
b	0.244	13.71
c	0.240	13.50
d	0.243	13.66

Table III-1) could be identically superimposed. Use of an integration step size of 0.001X resulted in a calculated depression angle within one percent of the depression angle for the 0.01X run. Utilization of a step size of greater than 0.01X is not recommended. Within this limitation the integrations can be carried out over 10X along the interface for a maximum field interval of 10X in the bottom. The 10X restriction is imposed by program array size. (For the J's specified, 0.5 degrees in the angle domain was slightly finer than the 0.01X field point step size in the depth domain and was the field point increment of choice).

IV. BEAM DEPRESSION ANGLE ANALYSIS

A simple relationship was sought between the physical parameters associated with the wedge system and the resultant beam.

Assume

$$\theta_{DC} \propto \beta^n \quad (\text{IV-1})$$

where θ_{DC} is the beam depression angle and β is the wedge angle. For each ρ_1/ρ_2 and C_1/C_2 , θ_{DC} vs β was graphed on log-log paper, and the slope (n) was measured. Results are listed in Table 1 of Appendix C, n equals 0.329.

Then θ_{DC} was found as a function of C_1/C_2 . The critical angle, θ_C , is an explicit function of C_1/C_2

$$\theta_C = \arccos (C_1/C_2) \quad \theta_C \text{ in radians} \quad (\text{IV-2})$$

It was then assumed

$$\theta_{DC} \propto \beta^{0.329} \theta_C^M \quad (\text{IV-3})$$

For any constant β , M was derived as the slope of a log-log graph of θ_{DC} vs θ_C . Graphs were analysed for β equal to 10, 6.92, 5, 2 and 1 degree, and ρ_1/ρ_2 ratios of 0.95, 0.90, 0.85, 0.70 and 0.50. Results are summarized in Table (2) of Appendix C, $M = 0.772$.

The same process was repeated for ascertaining dependence on ρ_1/ρ_2 . Log-log graphs were prepared by holding β and θ_C constant using the assumed relation

$$\theta_{DC} \propto \beta^{0.329} \theta_C^{0.772} (\rho_1/\rho_2)^q \quad (\text{IV-4})$$

Results are summarized in Table 3 of Appendix C. $q = -0.254$.

The proposed formula is now dependent on all physical properties, assuming the source angle is maintained at one-half β . To remove the proportionality sign, write

$$\theta_{DC} = K \beta^{0.329} \theta_C^{0.772} (\rho_1/\rho_2)^{-0.254} \quad (\text{IV-5})$$

and solve for K . From Table 4, Appendix C, $K = 17.22/\text{radian}$.

Therefore

$$\theta_{DC} = 17.22 \beta^{0.329} \theta_C^{0.772} (\rho_1/\rho_2)^{-0.254} \quad (\text{IV-6})$$

where β is in degrees and θ_C in radians.

θ_{DC} calculated from equation (IV-6) and the difference between θ_{DC} and θ_D from program WDGBTM (labeled $\Delta\theta_D$ and θ_{DW} respectively) are also tabulated on Table 4, Appendix C. $\Delta\theta_D$ varies from a minimum of 0.00 to a maximum of 1.25, with the average of the absolute values of $\Delta\theta_D$ equal to 0.36. Errors ranged from zero percent to 5.0 percent. Thus θ_{DC} gives a reasonable approximation of the beam depression angle.

A similar formula

$$\theta_{DA} = 17.1 \beta^{1/3} \theta_C^{0.773} (\rho_1/\rho_2)^{-0.27} \quad (IV-7)$$

was developed independently by A. Coppens (Reference 6). The minimum deviation of θ_{DA} from θ_{DW} is 0.01 and the maximum deviation is 1.56. The average of the absolute value of the deviation is 0.36. Both formulas are best for ρ_1/ρ_2 close to 1.0, and less accurate for ρ_1/ρ_2 equal to 0.5.

V. THE EFFECT OF ATTENUATION

As previewed in Section III-A.4 the pressure distribution as a function of η at $x = -9.0X$ demonstrated apparent interference effects (see figure V-1). As integration over the wedge interface to $4X$ encompasses dumping by the first and third mode, and the third mode dumps about the point $3X$ vice X for the first mode, it is logical to expect attenuation effects to reduce the contribution of the more distant mode. The initial Green's function approach, equation (II-9) did not consider attenuation of the acoustic wave in the bottom medium. An attenuation term

$$\exp\{-\alpha [(\xi - \xi')^2 + \eta^2]^{1/2}\} \quad (V-1)$$

was incorporated into equation (II-9) to yield

$$R(\vec{r}) = \eta/4 \exp\left(\frac{i\pi}{4}\right) \left(\frac{2k_2 X}{\pi}\right)^{1/2} \quad (V-2)$$

$$\int_0^{\infty} \frac{\tilde{p}(\xi') \exp\{-\alpha [(\xi - \xi')^2 + \eta^2]^{1/2}\} \exp\{i[L(\xi') - k_2 X [(\xi - \xi')^2 + \eta^2]^{1/2}]\}}{[(\xi - \xi')^2 + \eta^2]^{3/4}} d\xi'$$

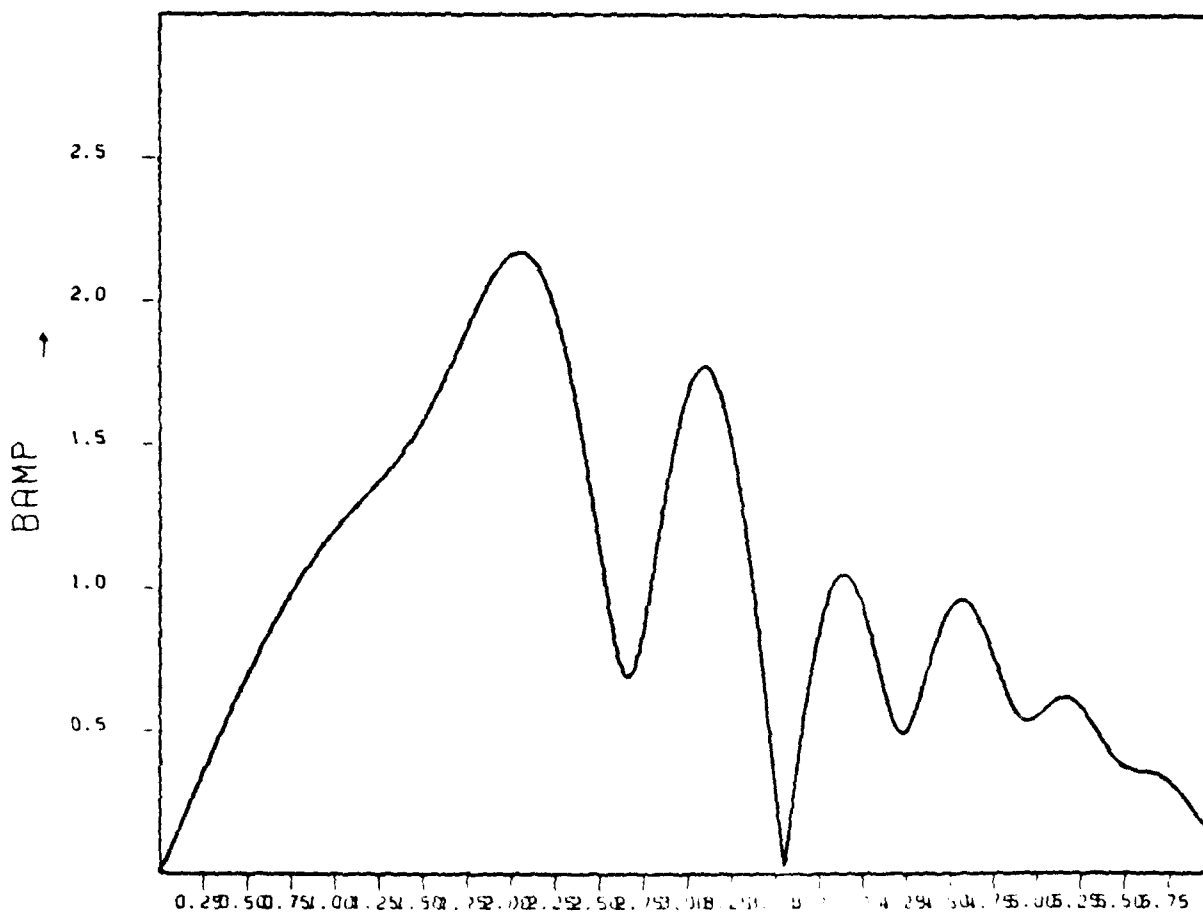
An appropriate code modification to the depth domain routine was made.

The attenuation coefficient α was an input variable to the program.

Two sets of comparison runs were made. The first set consisted of an unattenuated control ($\alpha = 0$) and an attenuated run with α equal 1.0, with the pressure amplitude measured down from the wedge apex. The second set consisted of an unattenuated control plus runs with α set to 1.0, 0.5, 0.1, and 0.01 and the pressure measured at $x = -9.0$ from the

WAVE	1.0000	1.00		
WAVE	2.0000	1.00	0.00	
WAVE	3.0000	1.00	0.00	
C1	1500.0000		FPN1	9.00
C2	1570.0000			
CHFC	0.9500			
BETA	2.50			
SMER	1.25			
SMED	999.00			
WINC	0.010			

Figure V-1



ETH

apex. All other input variables for both sets were held constant and are specified in Table (V-1). Integration over the wedge bottom was from the apex to four times the dump distance; this was to insure that contributions from the third mode (dumping about the point $x = 3X$) were included.

The results of set one are illustrated in Figure (V-2). Note that the second maxima is severely depressed in the attenuated case. The depression angle has been shifted slightly from 11.86 degrees for the unattenuated data to 11.31 degrees with α set to 1.0.

Figure V-3 is a plot of the results from set two. All curves have been normalized to the initial slope for $\alpha = 1.0$. The second and third maxima decrease with increasing α , and relatively more energy is contained in the first beam. The depression angle of the first beam decreases slightly with increasing α . These trends are summarized in Table (V-2).

Figures (V-4a) and (V-4b) illustrate the effects of attenuation on different contributing modes. For Figure (V-4a), the source was placed at $1/3 \beta$. With this source geometry, contributions from mode three are suppressed, and only the first and second mode contributed to the beam pattern in the bottom. For Figure (V-4b), the source was placed at $4/3 \beta$ and modes one and three propagate into the bottom medium. Both cases were run for $\alpha = 1.0, 0.5, 0.1$ and 0.0 . As could be expected, attenuation causes greater suppression of third mode interference contributions than of second mode contributions.

Table (V-1)

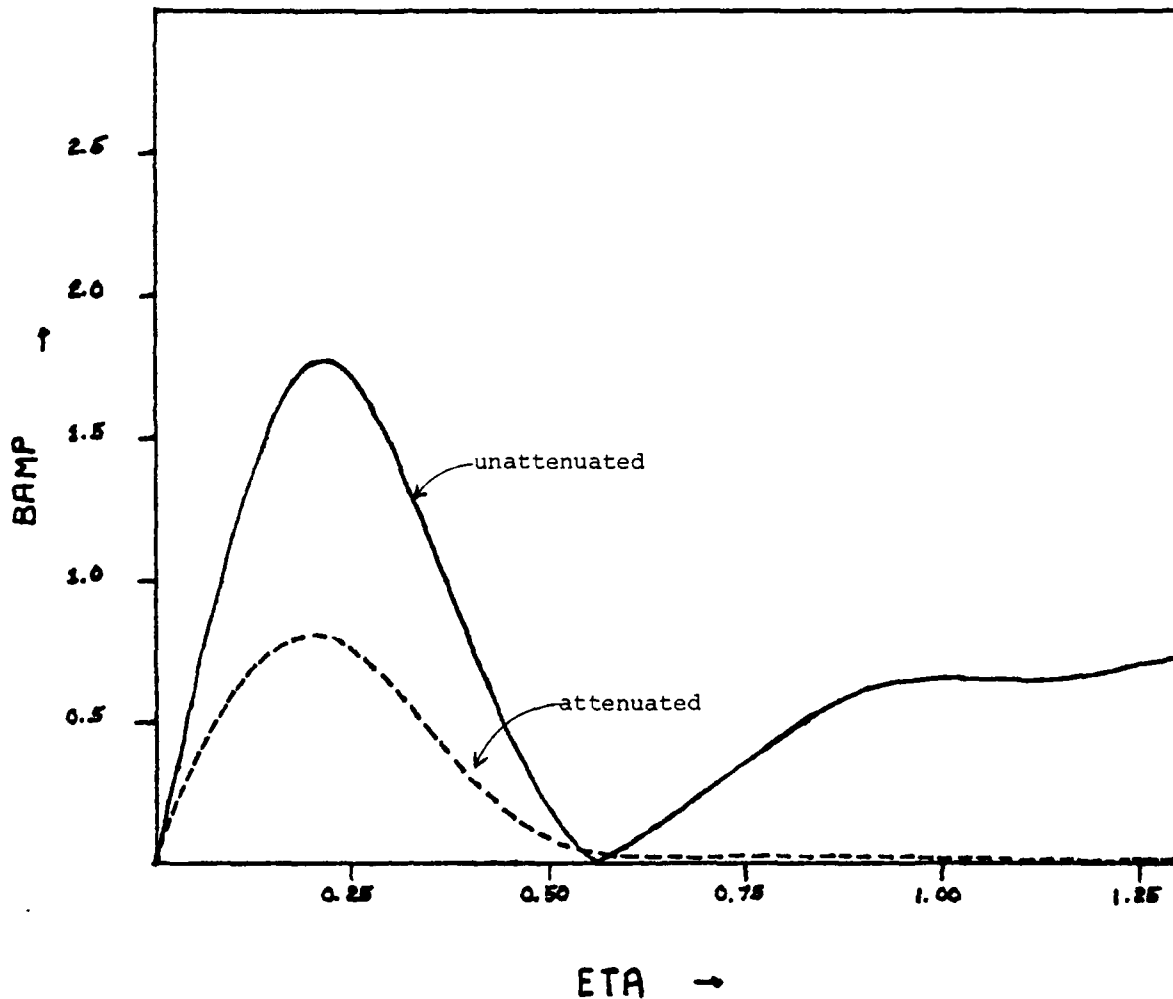
Input Specifications for Attenuation Runs

$C_1 = 1500.0$	$\rho_1 = 1.0$
$C_2 = 1578.947$	$\rho_2 = 1.052632$
$C_1/C_2 = 0.95$	$\rho_1/\rho_2 = 0.95$
$\beta = 5.0^\circ$	$f = 150000 \text{ hz}$
$X_2 = 4.0$	Source distance = infinite
$\delta = 2.5^\circ$	
$\alpha = 0.0, 0.01, 0.1, 0.5, 1.0$	

Refer to Figure II-1 for variable meaning.

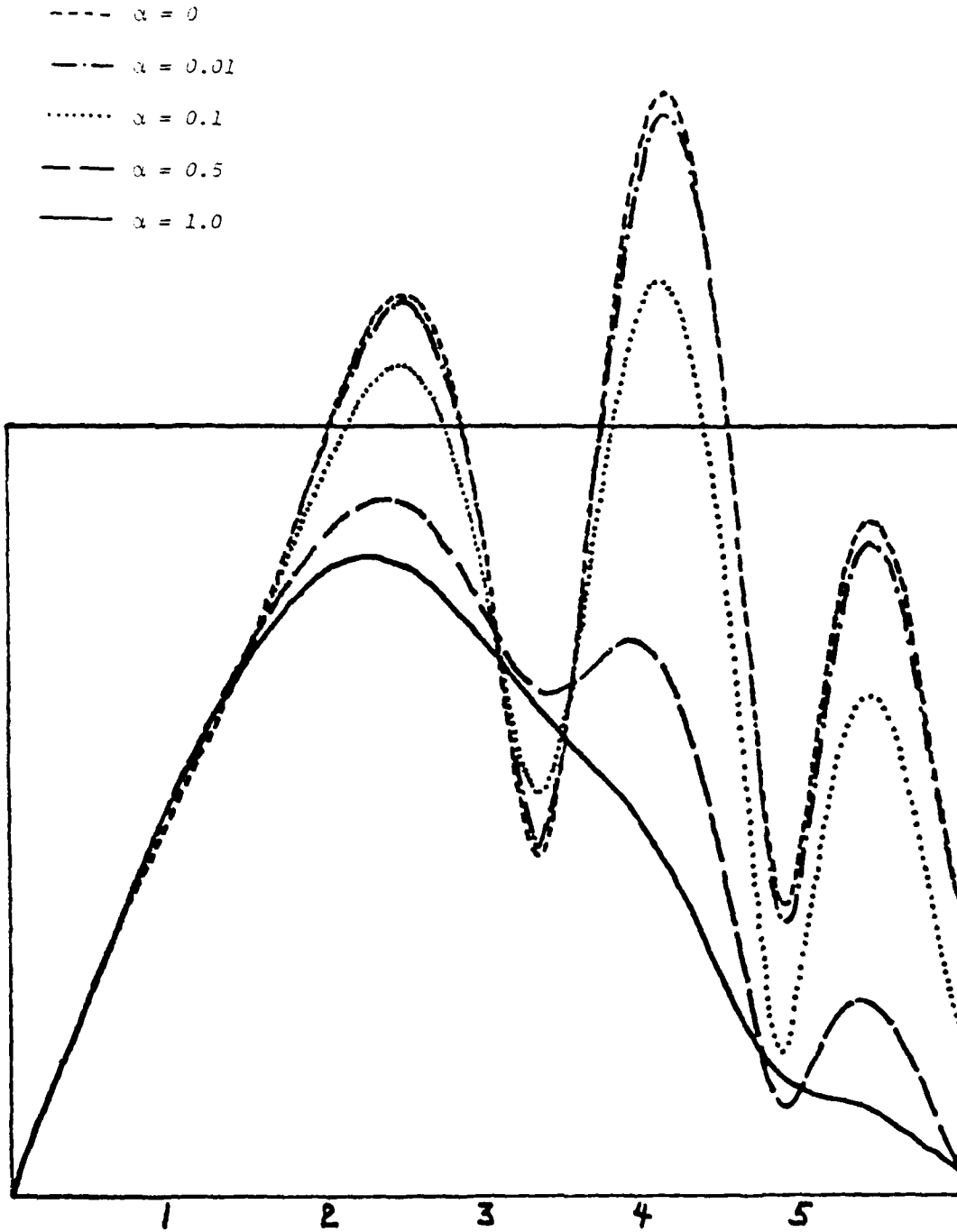
Figure V-2

Attenuated and unattenuated pressure amplitudes
down from the apex



ETA →

Figure V-3



$\eta \rightarrow$

TABLE V-2

α	γ	H_1	H_2	H_1/H_2
1.0	12.84	0.21×10^{-4}	NONE	
0.5	13.39	$.33 \times 10^{-2}$	$.26 \times 10^{-2}$	1.26
0.1	13.82	0.21	0.23	0.90
0.01	13.93	0.54	0.66	0.83
0.0	13.98	0.61	0.74	0.82

γ = depression angle of first maxima

H_1 = amplitude of first maxima

H_2 = amplitude of second maxima

RH01	0.8651	EBEG	0.0
RH02	1.1470	EEND	4.0
RRAT	0.8415	EINC	400.
C1	1500.0000	FPNT	0.00
C2	144.1984		
CRAT	0.8854		
BETA	1.53		
SACA	0.51		
SACD	999.00		
WINC	0.0.0		

Figure V-4a

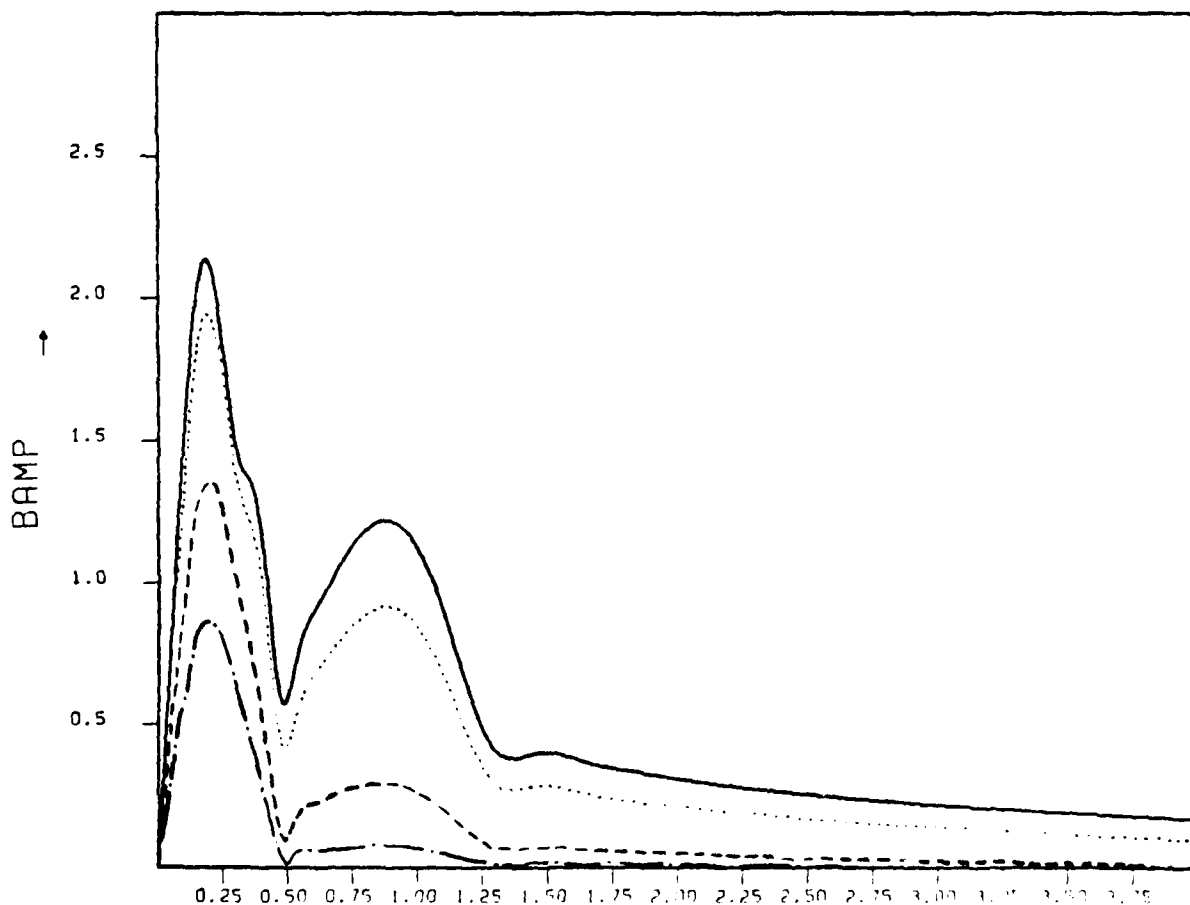
α

— 0.0

..... 0.1

- - - 0.5

- · - 1.0

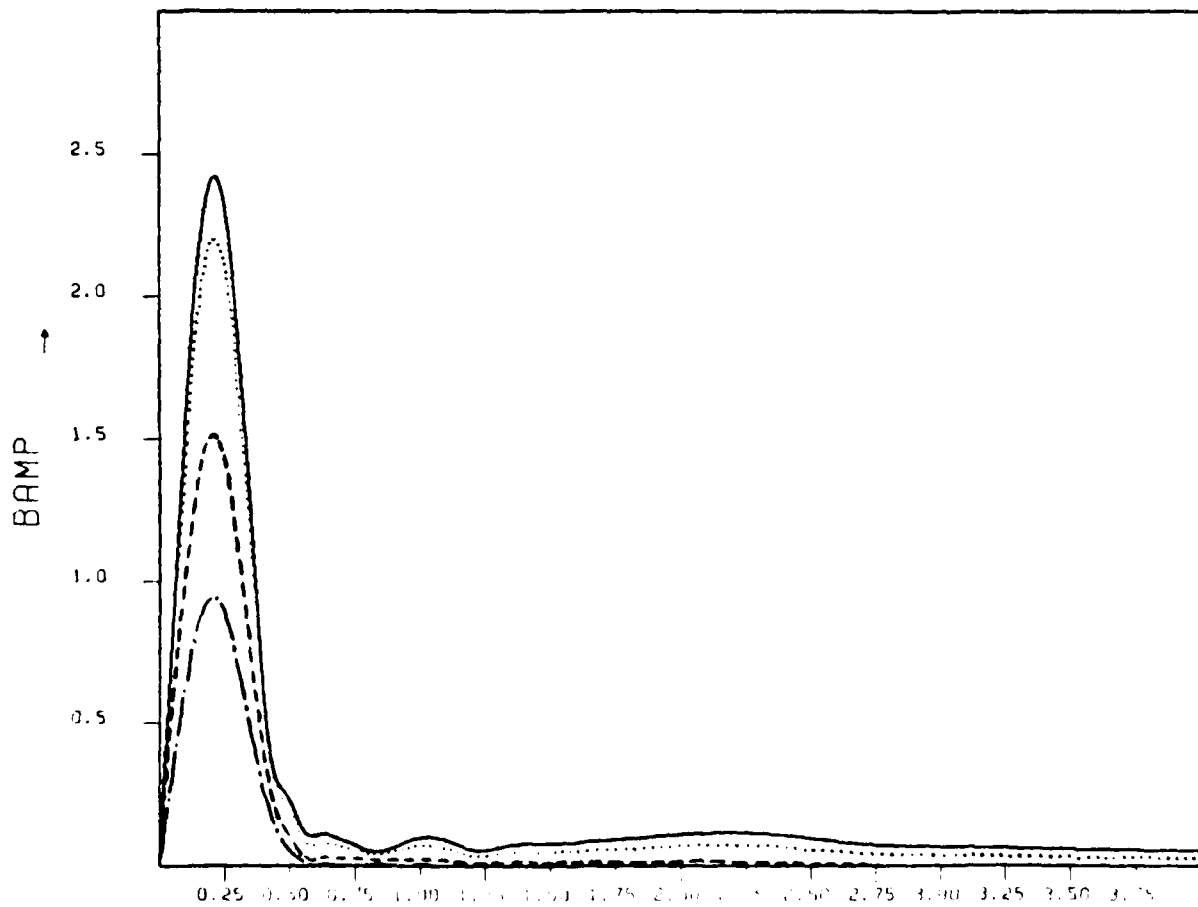


ETA →

RHO1	0.9651	EBEC	0.0
RHO2	1.1470	EEND	4.0
RAAT	0.8415	EINC	400.
C1	1500.0000	FPNT	0.00
C2	1694.1984		
CRAI	0.8854		
BETA	1.53		
SACA	0.76		
SACD	999.00		
WINC	0.010	α	

Figure V-4b

——— 0.0
 0.1
 - - - - 0.5
 - · - · 1.0



VI. ANALYSES OF FAR-FIELD INTERFERENCE PATTERNS

The results of the attenuation investigation were supportive of an interference explanation for the multiple beams. Subsequently, two analytical approaches were tried to predict the location of interference maxima and minima under unattenuated far-field conditions.

Both the Differential Change and Multiple Lambda analyses assume very simple processes. Both assume point sources of sound at the wedge interface located at X and $3X$. Realistically the sound sources behave as phased arrays propagating assymmetrically about the dumping points. Phase differences between the two dumping points are non-zero. The Differential Change formula becomes invalid as n approaches zero.

A. DIFFERENTIAL CHANGE APPROACH

Based on the simple geometric considerations illustrated in Fig (VI-1), the following nondimensional formula for the change in interference maxima locations was derived.

$$\Delta\eta = \frac{4 (C_2/C_1) \sin \theta_C \sin \beta}{\left\{ \frac{n}{[\eta^2 + (1-x)^2]^{1/2}} - \frac{n}{[\eta^2 + (3-x)^2]^{1/2}} \right\}} \quad (\text{VI-1})$$

Note that equation (VI-1) is dependent only on the wedge angle β and the sound speeds of the two fluid media.

A series of WDGTM executions were made for varying C_2/C_1 and β , with all other variables held constant. Table (VI-1) is a summary of one case, this case is typical of results observed. Results will be discussed in Section VI-C.

Source at $1/2 \beta$

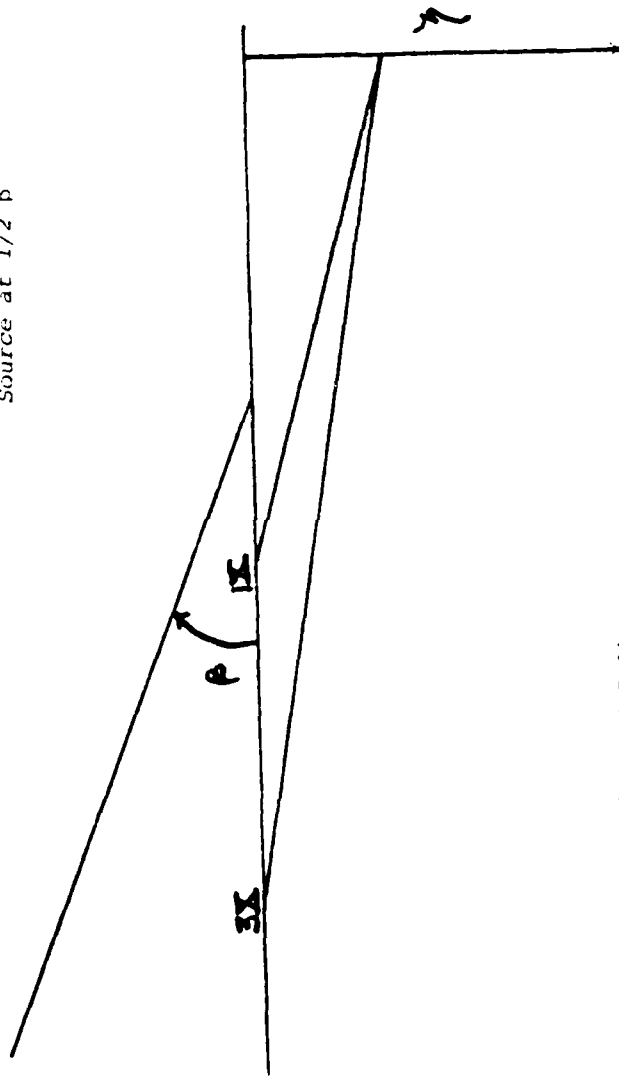


Figure (VI-1)

TABLE VI-1

Comparison of $\Delta\eta$ from equation (VI-1) and $\Delta\eta$ from WDGBTM results.

$$C_1/C_2 = 0.90$$

β	WDGBTM maxima locations	$\Delta\eta$ from WDGBTM	$\Delta\eta$ predicted
5.0	2.38	1.67	4.56
	4.05	1.92	3.04
	5.97		
2.5	2.33	1.40	2.33
	3.73	1.02	1.60
	4.75	0.91	1.38
	5.66		
1.25	2.08	1.22	1.29
	3.30	0.66	0.88
	3.96	0.49	0.77
	4.45		

B. MULTIPLE LAMBDA APPROACH

The second interference pattern calculation also utilizes the geometry of Fig. (VI-1). For each C_1/C_2 ratio and beta combination, X and λ_2 (the wavelength in the bottom media) was calculated according to the equations

$$X = \frac{C_1}{4f \sin \theta_C \sin \beta} \quad (\text{VI-2})$$

and

$$\lambda_2 = C_2/f \quad (\text{VI-3})$$

where f is 150kHz. The ratio $\lambda_2/X = Q$ is then found and used in the relation

$$NQ = [n^2 + (3-X)^2]^{1/2} - [n^2 + (1-X)^2]^{1/2} \quad (\text{VI-4})$$

where N is any positive integer. Note that use of equation (VI-4) assumes zero initial phase.

To find the expected maxima and minima, a plot of n vs N was made for each Q . Maxima are expected for each integer N , minima for each $N + 1/2$. Table (VI-2) summarizes one case of maxima and minima expected as a result of equation (VI-4) versus maxima and minima observed on WDGTM output.

C. SUMMARY OF ANALYSES

Predictions from both simple analyses are of the same order of magnitude as WDGTM results. The number of maxima and minima are

TABLE VI-2

OCCURRENCE OF MAXIMA AND MINIMA

$$C_1/C_2 = 0.85$$

β (DEG)	η MAX WDGBTM	η MAX EXP	η MIN WDGBTM	η MIN EXP
5	4.03	2.68 6.34	5.39	4.70
2.5	3.59 5.00	2.59 4.70	4.55 5.66	3.76 5.56
1.25	2.02 2.85 3.81 5.03 5.62	2.53 3.72 4.68 5.54	2.49 3.56 4.77 5.40 5.90	1.76 3.16 4.25 5.13 5.94

consistently predicted as increasing with decreasing wedge angle. Occasional unexplainable phenomena, such as increasing WDGTM $\Delta\eta$ for the $\beta = 5$ on Table (VI-1) and the extra WDGTM maxima for $\beta = 5$ Table (VI-2), were observed, but results are generally supportive of an interference effect occurring.

VII. COMPARISONS

A. EXPERIMENTAL DATA

Netzorg (Reference 4) measured pressure amplitude as a function of angle for fixed radii in four media below a wedge shaped upper layer. The angle domain version of WDGTM was used to compare calculated pressure patterns with measured patterns.

All runs utilized finite source input. For each of Netzorg's four experimental cases, corresponding computer runs were made with the source distance set to 100X, 10X and SX, where S corresponds to the actual source distance used in the tank. In all cases S was less than 10X. The integrations were carried out over one dumping distance and four dumping distances for each source.

Figure (VII-1) is typical of pressure amplitudes in the bottom. The source in the experiment was only about 40 wavelengths from the dumping point, even at 10X the source is approximately 110 wavelengths away. Since the method of images calculations utilized to determine the pressure amplitude along the wedge bottom assume a plane wave, it is questionable that this program can produce a realistic comparison with the measured data.

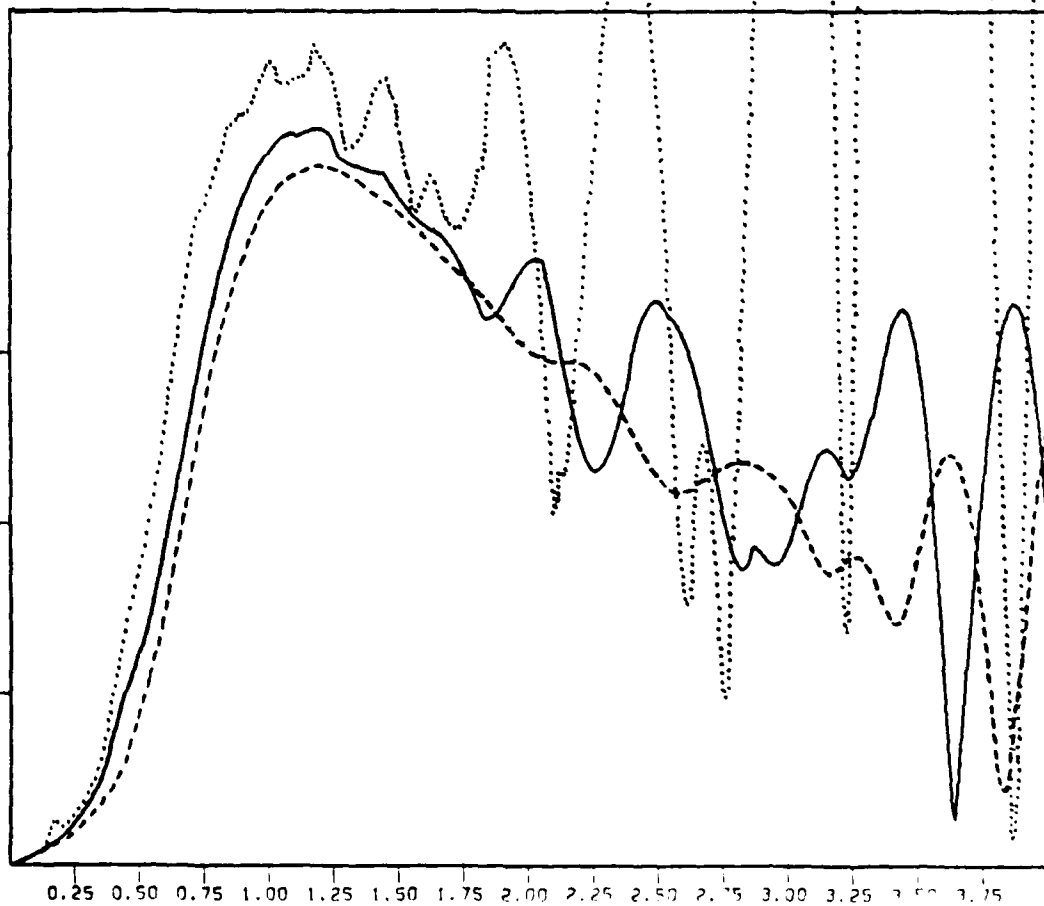
Figures (VII-2) and (VII-3) illustrate the wide variation in results obtained depending on the source distance and integration interval. Calculations demonstrate expected trends, i.e., pressure amplitude increases with decreasing radius distance (ρ), with closing source

Figure VII-1

RHO1	0.9651
RHO2	1.1470
RAAT	0.8415
C1	1500.0000
C2	1694.1984
CAAT	0.8854
BETA	2.51
SRCA	1.25
SACD	1.23
WINC	0.010

-- source at 100X
— source at 10X
..... source at 3.5X

WAMP



CAPX

Figure VII-2

PH01	0.9051	SCMA	1.21	
PH02	1.1470	TBEU	0.00	
ARAT	0.8415	TEND	45.00	
C1	1500.0000	TINC	0.500	
C2	1694.1984			Source Distance
CAAT	0.8854			X_2
BETA	2.51			
SACA	1.25			
SACD	1.23			
WINC	0.010			

---	3.5	1
- - -	10	1
- - -	100	1
.....	3.5	4
———	10	4
- - -	100	4

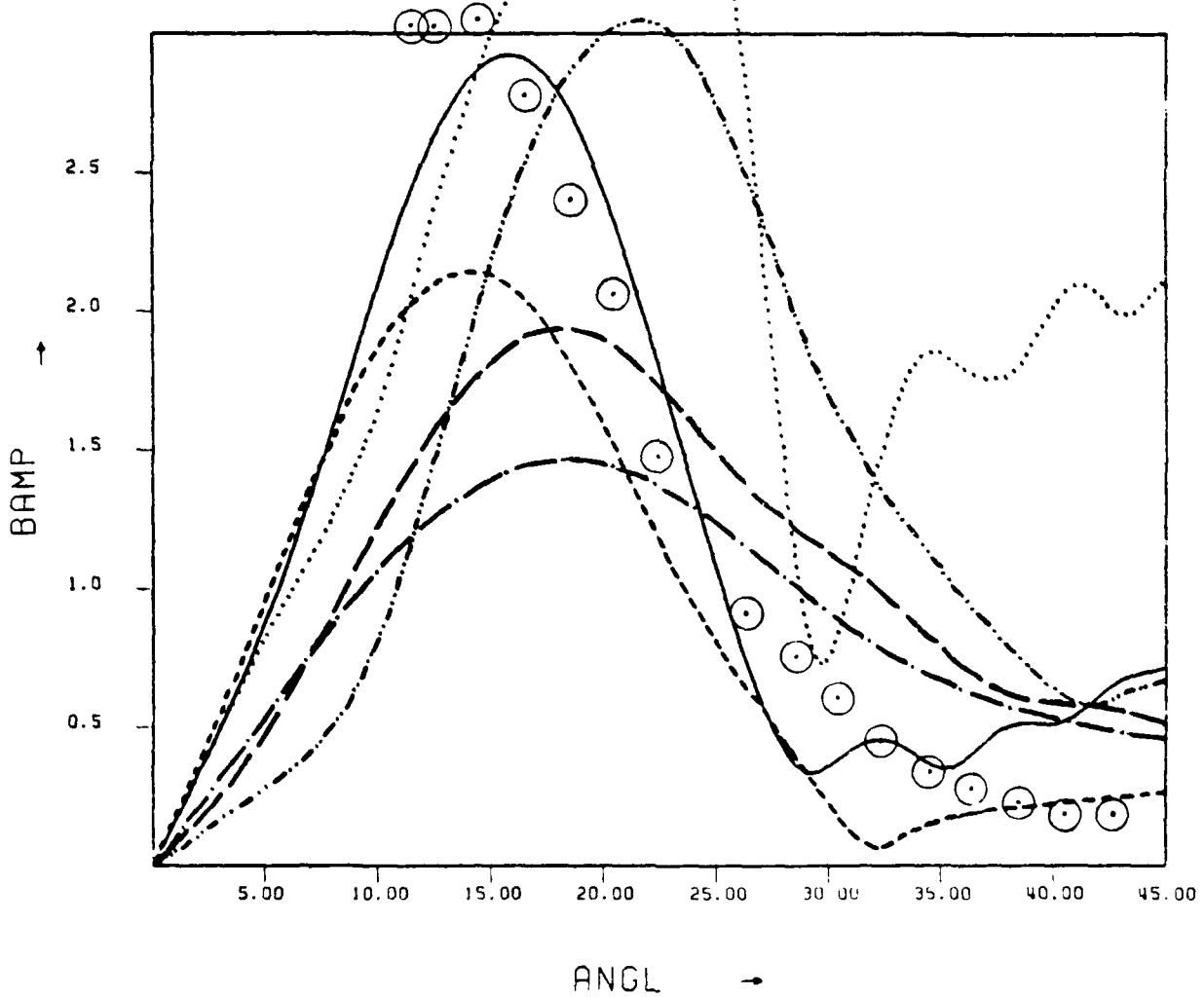
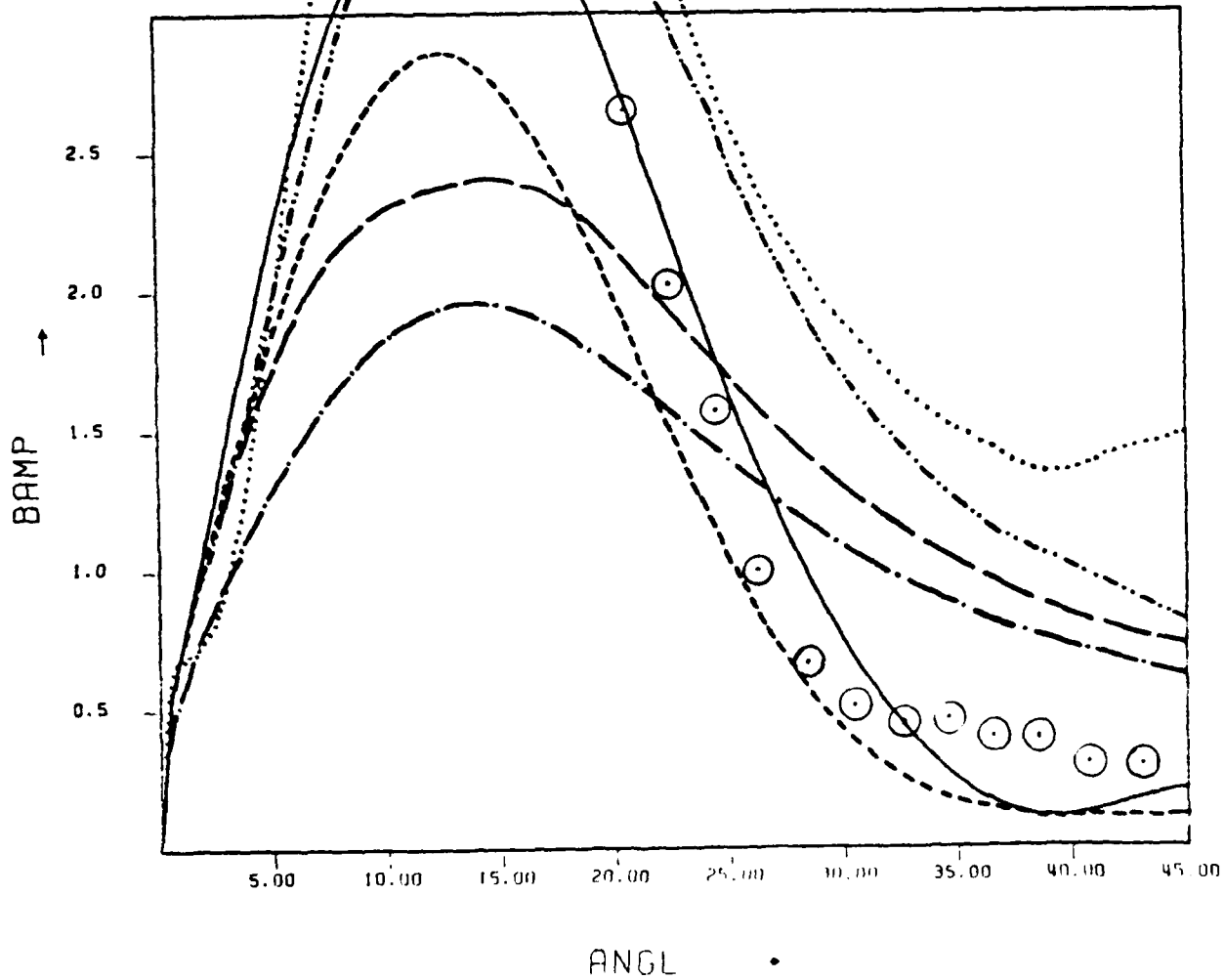


Figure VII-3

AM01	0.1051	SUMR	1.63
AM02	1.1970	TRIG	9.00
AMAT	0.8415	TEND	45.00
C1	1900.0000	TINC	0.500
C2	1694.1984		
CRAT	0.8054		
BETA	2.51		
SACA	1.25		
SACD	1.23		
WINC	0.010		

Source Distance	X ₂
--- 3.5	1
— 10	1
- - - 100	1
..... 3.5	4
— 10	4
- - - 100	4



distance, and as a function of an increasing integration interval over the wedge bottom. The beam depression angle becomes more shallow as source distance increases. Results at very low angles are not considered valid because of assumptions inherent in the WDGTM model.

The circled data points of Figures (VII-2) and (VII-3) are a transposition of data from Reference 4. Since absolute amplitude was not specified in the reference, rescaling to the case most near to the actual conditions was performed, i.e., the data were normalized to the case $X_2 = 1.0$ and $srcd = 3.5X$. The shape of the experimental curve and the apparent beam depression angle are the only valid comparisons that should be made.

Comparisons of WDGTM beam depression angles with measured values gave varying results. Measured depression angles decreased with increasing σ , WDGTM depression angles increased. WDGTM results were within 6 percent to 62 percent of measured values. No correlation was observed between prediction accuracy and source distance or prediction accuracy and the interface integration interval. All measured beam depression angles in Reference 8 were for the two shortest radii of each case, the three longer radii could only be qualitatively compared with graphical data. Possible attenuation effects were not considered. Numerical results are contained in Appendix D.

B. PARABOLIC EQUATION APPROACH

Kuperman (Reference 9) describes a Parabolic Equation model result for a wedge problem, and compares these results to one set of data extracted from Reference 10.

The Parabolic Equation results show very well defined beams in the bottom, with no apparent interference effects as observed in comparable WDGBTM runs (Figures V-1 and V-4). Kuperman used an attenuation factor of 0.5dB per wave length, a value equivalent to an input of 0.984 Nepers/X for WDGBTM. It should be noted in Figure V-4 that there is complete separation of the beams for WDGBTM when $\alpha = 1.0$.

Kuperman states agreement within 20 percent of a measured data set of Reference 10. This measured data is identical to Case 1 of Appendix D. WDGBTM accuracy for this case varied from zero percent error to 22 percent error with error calculated according to the formula

$$100 \times \frac{(\text{observed data}) - (\text{measured data})}{\text{measured data}}$$

VIII. CONCLUSIONS

1. Program WDGBTM offers a fast, efficient means of analysing the behavior of sound in a fast bottom underlying a wedge-shaped medium. Effects of attenuation are easily demonstrated, and modes desired are readily controlled by specification of source location and integration interval selected. Results of the model are not inconsistent with current measured data, and illuminate observations generated by the Parabolic Equation model.

2. A simple equation for calculating beam depression angle as a function of wedge characteristics was obtained which offers sufficient accuracy for use as a rapid analytical tool. The equation was derived for acoustic sources placed at one-half the wedge angle. Accuracy degrades when the wedge medium density to bottom medium density ratio decreases to 0.5.

3. It is recommended that further experiments be performed that would further test the model. New experiments should be sufficiently large scale to ensure plane waves at the wedge bottom interface and to satisfy the $k_2 |\vec{r} - \vec{r}'| \gg 2\pi$ criteria. Bottom media that reasonably model real ocean fluid-like bottom are also recommended.

APPENDIX A

SAMPLE INPUT SET

INFINITE SOURCE INPUT	1500.0	1578.94737	1.00000	1.0526315	5.0
0.0	4.0		150000.	2.5	
DEPTH DOMAIN INPUT	0.0	6.00		-9.0	600
0.0	100			-9.0	600
DEPTH DOMAIN INPUT	0.0	1628.0	0.98666	1.08742	2.66
0.0	4.0		150000.	1.33	
INFINITE SOURCE INPUT	10.0				
ANGLE DOMAIN INPUT	1.31	45.0	0.5		
0.0	0.0				
ANGLE DOMAIN INPUT	0.90	45.0	0.5		
0.0	0.0				
DEPTH DOMAIN INPUT	0.0	1.3		0.0	130
0.0	0.000				

Key on following page

Key to Input Set

Line #	Value	Format
1	Litoral string for infinite source option	A4
2	$C_1, C_2, \rho_1, \rho_2, \beta,$	5F15.0
3	$\xi_1, \xi_2, f, \delta,$	4F15.0
4	number of increments for $\xi_2 - \xi_1$	I4
5	Litoral string for depth domain option	A4
6	$\eta_1, \eta_2, \xi,$ number of increments between $\eta_2 - \eta_1$	3(F15.7, 5X), I5
7	α	F15.7
8	see line 5	
9	see line 6	
10	see line 7	
11	Litoral string for finite source option	A4
12	see line 2	
13	see line 3	
14	see line 4	
15	source distance in x	F15.7
16	Litoral string for angle domain input	A4
17	$\sigma, \theta_{k1}, \theta_{k2}, \Delta\theta_k$	4(F5.2, 5X)
18	see line 16	
19	see line 17	
20	see line 5	
21	see line 6	
22	see line 7	

APPENDIX B

$$\rho_1/\rho_2 = .9500$$

c_1/c_2	θ (DEG)	Δ DEPRESSION	BANDWIDTH	RATIO
0.95	10.00	15.64	22.93	0.68
	6.92	13.50	19.08	0.71
	5.00	12.02	17.48	0.69
	2.00	9.09	12.19	0.74
	1.00	7.58	9.54	0.79
0.90	10.00	19.80	29.07	0.68
	6.92	17.74	26.66	0.67
	5.00	16.33	21.85	0.75
	2.00	11.31	16.20	0.70
	1.00	9.09	12.73	0.71
0.85	10.00	24.37	36.02	0.68
	6.92	21.16	31.05	0.68
	5.00	18.42	26.61	0.69
	2.00	13.50	19.45	0.69
	1.00	11.31	14.36	0.79

Δ DEPRESSION = ARCTAN(Z/X) AT MAX AMPLITUDE

$$\text{BANDWIDTH} = \text{ARCTAN}(Z/X_{1u} - Z/X_{1L})$$

$$\text{MEAM } \frac{\Delta_D}{\text{BW}} = .73$$

$$\rho_1/\rho_2 = .9000$$

C_1/C_2	β (DEG)	Δ DEPRESSION	BANDWIDTH	RATIO
0.95	10.00	15.64	22.68	0.69
	6.92	13.49	19.08	0.71
	5.00	12.02	17.43	0.69
	2.00	9.09	12.19	0.74
	1.00	7.58	9.54	0.79
0.90	10.00	19.80	29.77	0.66
	6.92	18.42	26.75	0.69
	5.00	16.33	21.80	0.75
	2.00	12.02	15.53	0.77
	1.00	9.09	12.90	0.70
0.85	10.00	24.37	36.28	0.67
	6.92	21.16	31.22	0.68
	5.00	18.42	26.93	0.68
	2.00	13.50	19.70	0.68
	1.00	11.31	14.95	0.76

$$\rho_1/\rho_2 = .8000$$

c_1/c_2	β (DEG)	Δ DEPRESSION	BANDWIDTH	RATIO
0.95	10.00	16.33	22.39	.73
	6.92	14.20	19.19	.74
	5.00	12.79	17.33	.74
	2.00	9.82	12.13	.81
	1.00	7.58	9.87	.77
0.90	10.00	20.46	31.00	.66
	6.92	19.14	26.93	.71
	5.00	16.33	21.90	.74
	2.00	12.02	16.38	.73
	1.00	9.82	12.95	.76
0.85	10.00	25.03	36.94	.68
	6.92	22.44	31.72	.71
	5.00	19.14	27.65	.69
	2.00	14.20	20.10	.71
	1.00	11.31	15.32	.74

$$\rho_1/\rho_2 = .7000$$

c_1/c_2	β (DEG)	Δ DEPRESSION	BANDWIDTH	RATIO
0.95	10.00	17.07	22.24	.77
	6.92	14.20	19.34	.73
	5.00	13.50	17.17	.79
	2.00	9.82	12.13	.81
	1.00	8.36	9.65	.87
0.90	10.00	21.16	32.66	.65
	6.92	19.80	26.93	.74
	5.00	17.07	22.00	.78
	2.00	12.79	16.59	.77
	1.00	10.59	13.12	.81
0.85	10.00	25.64	37.88	.68
	6.92	23.12	32.29	.72
	5.00	19.80	28.72	.69
	2.00	14.95	20.56	.73
	1.00	12.02	15.85	.76

$$\rho_1/\rho_2 = .5000$$

C_1/C_2	β (DEG)	Δ DEPRESSION	BANDWIDTH	RATIO
0.95	10.00	17.74	22.73	0.78
	6.92	15.64	19.80	0.79
	5.00	14.95	16.91	0.88
	2.00	9.82	12.68	0.77
	1.00	9.09	10.26	0.89
0.90	10.00	22.44	35.53	0.63
	6.92	21.80	27.16	0.80
	5.00	18.42	23.70	0.78
	2.00	14.20	17.64	0.80
	1.00	11.31	13.87	0.82
0.85	10.00	27.20	42.80	0.64
	6.92	25.03	34.33	0.73
	5.00	21.80	31.00	0.70
	2.00	17.07	21.40	0.78
	1.00	13.50	16.91	0.80

APPENDIX C

TABLE 1

ρ_1/ρ_2	C_1/C_2	n
0.95	0.95	0.318
0.95	0.90	0.347
0.95	0.85	0.341
0.90	0.95	0.318
0.90	0.90	0.335
0.90	0.85	0.339
0.80	0.95	0.328
0.80	0.90	0.328
0.80	0.85	0.343
0.70	0.95	0.318
0.70	0.90	0.303
0.70	0.85	0.343
0.50	0.95	0.339
0.50	0.90	0.314
0.50	0.85	0.314

Average n = 0.329

Standard (N-1) deviation = 0.014

(N-1) Variance = 0.002

TABLE 2

σ_1/σ_2	S	M
0.95	10.0	0.834
0.95	6.92	0.845
0.95	5.00	0.783
0.95	2.00	0.739
0.95	1.00	0.752
0.90	10.0	0.815
0.90	6.92	0.818
0.90	5.00	0.782
0.90	2.00	0.735
0.90	1.00	0.720
0.80	10.0	0.791
0.80	6.92	0.839
0.80	5.00	0.767
0.80	2.00	0.686
0.30	1.00	0.750
0.70	10.0	0.730
0.70	6.92	0.892
0.70	5.00	0.701
0.70	2.00	0.778
0.70	1.00	0.697
0.50	10.0	0.786
0.50	6.92	0.881
0.50	5.00	0.693
0.50	2.00	1.051*
0.50	1.00	0.722

*Rejected data point	Average M	= 0.772
	Standard (N-1) deviation	= 0.059
	Variance	= 0.003

Table 3

C_1/C_2	$\theta_c(\text{rad})$	β	q
0.95	0.318	10.0	-0.217
0.95	0.318	6.92	-0.285
0.95	0.318	5.00	-0.257
0.95	0.318	2.00	-0.215
0.95	0.318	1.00	-0.283
0.90	0.451	10.0	-0.183
0.90	0.451	6.92	-0.232
0.90	0.451	5.00	-0.211
0.90	0.451	2.00	-0.300
0.90	0.451	1.00	-0.317
0.85	0.555	10.0	-0.162
0.85	0.555	6.92	-0.242
0.85	0.555	5.00	-0.255
0.85	0.555	2.00	-0.362
0.85	0.555	1.00	-0.283

Average $q = -0.254$

standard (N-1) deviation = 0.053

variation = 0.0028

Table 4

ρ_1/ρ_2	c_1/c_2	β	θ_{DW}	K	θ_{DC}	$\Delta\theta_D$
0.95	0.95	10.0	15.64	17.54	15.35	-0.29
0.95	0.95	6.92	13.50	17.08	13.60	0.10
0.95	0.95	5.00	12.02	16.94	12.22	0.20
0.95	0.95	2.00	9.09	17.32	9.04	-0.05
0.95	0.95	1.00	7.58	18.14	7.20	-0.38
0.95	0.90	10.0	19.80	16.94	20.12	0.32
0.95	0.90	6.92	17.74	17.13	17.83	0.09
0.95	0.90	5.00	16.33	17.55	16.02	-0.31
0.95	0.90	2.00	11.31	16.43	11.85	0.54
0.95	0.90	1.00	9.09	16.59	9.43	0.32
0.95	0.85	10.0	24.37	17.77	23.61	-0.76
0.95	0.85	6.92	21.16	17.42	20.92	-0.24
0.95	0.85	5.00	18.42	16.87	18.80	0.38
0.95	0.85	2.00	13.50	16.72	13.91	0.41
0.95	0.85	1.00	11.31	17.59	11.07	-0.24
0.90	0.95	10.0	15.64	17.31	15.56	-0.08
0.90	0.95	6.92	13.49	16.85	13.79	0.30
0.90	0.95	5.00	12.02	16.71	12.39	0.37
0.90	0.95	2.00	9.09	17.08	9.16	0.07
0.90	0.95	1.00	7.58	17.89	7.30	-0.28
0.90	0.90	10.0	19.80	16.71	20.40	0.60
0.90	0.90	6.92	18.42	17.55	18.08	-0.34
0.90	0.90	5.00	16.33	17.31	16.24	-0.09
0.90	0.90	2.00	12.02	17.23	12.02	0.00
0.90	0.90	1.00	9.09	16.36	9.56	0.47
0.90	0.85	10.0	24.37	17.53	23.94	-0.43
0.90	0.85	6.92	21.16	17.18	21.21	-0.05
0.90	0.85	5.00	18.42	16.64	19.06	0.64
0.90	0.85	2.00	13.50	16.49	14.10	0.60
0.90	0.85	1.00	11.31	17.35	11.22	-0.09

Table 4 (continued)

ρ_1/ρ_2	c_1/c_2		θ_{DW}	K	θ_{DC}	$\Delta\theta_D$
0.80	0.95	10.0	16.33	17.54	16.03	-0.30
0.80	0.95	6.92	14.20	17.21	14.20	0.00
0.80	0.95	5.00	12.79	17.25	12.76	-0.03
0.80	0.95	2.00	9.82	17.91	9.44	-0.38
0.80	0.95	1.00	7.58	17.36	7.52	-0.06
0.80	0.90	10.0	20.46	16.76	21.02	0.00
0.80	0.90	6.92	19.14	17.70	18.62	-0.52
0.80	0.90	5.00	16.33	16.80	16.54	0.21
0.80	0.90	2.00	12.02	16.72	12.38	0.36
0.80	0.90	1.00	9.82	17.16	9.86	0.04
0.80	0.85	10.0	25.03	17.47	24.67	-0.36
0.80	0.85	6.92	22.44	17.68	21.85	-0.59
0.80	0.85	5.00	19.14	16.78	19.64	0.50
0.80	0.85	2.00	14.20	16.83	14.53	0.33
0.80	0.85	1.00	11.31	16.84	11.56	0.25
0.70	0.95	10.0	17.07	17.72	16.59	-0.48
0.70	0.95	6.92	14.20	16.64	14.70	0.50
0.70	0.95	5.00	13.50	17.60	13.21	-0.29
0.70	0.95	2.00	9.82	17.31	9.77	0.05
0.70	0.95	1.00	8.36	18.51	7.78	-0.58
0.70	0.90	10.0	21.16	16.75	21.75	0.59
0.70	0.90	6.92	19.80	17.70	19.27	-0.53
0.70	0.90	5.00	17.07	16.98	17.31	0.24
0.70	0.90	2.00	12.79	17.20	12.81	0.02
0.70	0.90	1.00	10.59	17.88	10.20	-0.39
0.70	0.85	10.0	25.64	17.30	25.52	-0.12
0.70	0.85	6.92	23.12	17.61	22.61	-0.51
0.70	0.85	5.00	19.80	16.78	20.32	0.52
0.70	0.85	2.00	14.95	17.13	15.03	0.08
0.70	0.85	1.00	12.02	17.30	11.96	0.06

Table 4 (continued)

ρ_1/ρ_2	c_1/c_2	β	θ_{DW}	K	θ_{DC}	$\Delta\theta_D$
0.50	0.95	10.0	17.74	16.91	18.07	0.33
0.50	0.95	6.92	15.64	16.82	16.01	0.37
0.50	0.95	5.00	14.95	17.90	14.38	-0.57
0.50	0.95	2.00	9.82	15.89	10.64	0.82
0.50	0.95	1.00	9.09	18.48	8.47	-0.62
0.50	0.90	10.0	22.44	16.31	23.69	1.25
0.50	0.90	6.92	21.80	17.89	20.99	-0.81
0.50	0.90	5.00	18.42	16.82	18.86	0.44
0.50	0.90	2.00	14.20	17.53	13.95	-0.25
0.50	0.90	1.00	11.31	17.54	11.10	-0.21
0.50	0.85	10.0	27.20	16.85	27.80	0.60
0.50	0.85	6.92	25.03	17.50	24.62	-0.41
0.50	0.85	5.00	21.80	16.96	22.13	0.33
0.50	0.85	2.00	17.07	17.96	16.37	-0.70
0.50	0.85	1.00	13.50	17.84	13.03	-0.42

Average K = 17.22

Standard (N-1) deviation = 0.51

(N-1) variance = 0.26

APPENDIX D

Case 1

x_s	x_{max}	σ	ϑ_{DW}	BW_w	Measured ϑ_D	Measured BW	
2.12	1	1.60	24.5	10.33			
	1	1.10	22.0	11.34			
	1	0.83	19.0	12.88			
	1	0.66	16.5	15.27	13.5	15	
	1	0.55	14.5	18.38	15.5	15	
	4	1.60	23.0	6.28			
	4	1.10	22.1	8.19			
	4	0.83	19.0	12.20			
	4	0.66	16.0	13.88	13.5	15	
	4	0.55	15.0	14.71	15.5	15	
	10	1	1.60	17.5	19.26		
		1	1.10	16.0	20.29		
		1	0.83	16.5	20.90		
		1	0.66	13.5	22.41	13.5	15
1		0.55	12.5	23.19	15.5	15	
4		1.60	16.0	13.85			
4		1.10	13.5	14.96			
4		0.83	13.0	15.36			
4		0.66	12.0	15.67	13.5	15	
4		0.55	12.0	16.33	15.5	15	
100		1	1.60	16.5	22.68		
		1	1.10	15.5	23.27		
		1	0.83	14.0	23.91		
		1	0.66	13.0	24.81	13.5	15
	1	0.55	11.5	26.00	15.5	15	
	4	1.60	12.0	15.18			
	4	1.10	12.0	14.93			
	4	0.83	11.5	15.25			
	4	0.66	11.0	15.94	13.5	15	
	4	0.55	10.5	16.95	15.5	15	

$$C_1/C_2 = 0.8854$$

$$\rho_1/\rho_2 = 0.8415$$

$$\beta = 1.52^\circ$$

Case 2

x_s	x_{max}	σ	a_{DW}	BW_w	Measured θ_D	Measured BW	
3.50	1	1.21	21.5	18.31			
	1	0.83	17.5	20.60			
	1	0.63	14.5	24.19			
	1	0.50	12.5	26.62	13.3	15.5	
	1	0.42	10.0	27.41	14.0	14.38	
	4	1.21	22.5	15.04			
	4	0.83	16.0	19.24			
	4	0.63	14.0	19.32			
	4	0.50	13.0	20.34	13.3	15.5	
	4	0.42	11.5	22.75	14.0	14.38	
	10	1	1.21	18.0	24.19		
		1	0.83	14.5	25.13		
1		0.63	14.5	28.69			
1		0.50	12.0	29.56	13.3	15.5	
1		0.42	9.5	30.18	14.0	14.38	
4		1.21	16.0	16.07			
4		0.83	14.0	17.97			
4		0.63	14.0	20.43			
4		0.50	12.5	22.70	13.3	15.5	
4		0.42	11.5	24.96	14.0	14.38	
100		1	1.21	18.5	26.64		
		1	0.83	16.0	27.72		
	1	0.63	14.0	29.29			
	1	0.50	11.5	30.89	13.3	15.5	
	1	0.42	9.5	31.55	14.0	14.38	
	4	1.21	14.0	17.65			
	4	0.83	13.5	17.94			
	4	0.63	12.5	19.37			
	4	0.50	11.5	21.69	13.3	15.5	
	4	0.42	11.0	23.80	14.0	14.38	

$$C_1/C_2 = 0.8854$$

$$\rho_1/\rho_2 = 0.8415$$

$$\beta = 2.51^\circ$$

Case 3

x_s	x_{max}	σ	θ_{DW}	BW_w	Measured θ_D	Measured θ_W	
6.15	1	0.58	13.5	30.13			
	1	0.40	8.0	32.50			
	1	0.30	7.0	33.47			
	1	0.24	7.0	35.75	16.4	25.5	
	1	0.20	7.5	38.56	20	21	
	4	0.58	14.0	24.62			
	4	0.40	10.5	30.65			
	4	0.30	10.5	35.06			
	4	0.24	10.5	41.51	16.4	25.5	
	4	0.20	11.0		20	21	
	10	1	0.58	14.5	24.60		
		1	0.40	12.5	30.47		
1		0.30	11.5	35.89			
1		0.24	11.5	41.62	16.4	25.5	
1		0.20	12.0		20	21	
4		0.58	14.5	33.94			
4		0.40	9.0	35.10			
4		0.30	7.5	36.04			
4		0.24	7.5	38.04	16.4	25.5	
4		0.20	7.5	40.78	21	21	
100		1	0.58	14.0	35.19		
		1	0.40	8.5	35.83		
	1	0.30	7.0	36.47			
	1	0.24	7.0	38.37	16.4	25.5	
	1	0.20	7.5	40.82	20	21	
	4	0.58	13.5	25.14			
	4	0.40	11.0	30.32			
	4	0.30	10.0	35.38			
	4	0.24	10.0	40.97	16.4	25.5	
	4	0.20	11.0	35.02	20	21	

$$C_1/C_2 = 0.8854$$

$$\rho_1/\rho_2 = 0.8415$$

$$\beta = 4.41^\circ$$

Case 4

x_s	x_{max}	σ	θ_{DW}	BW_w	Measured θ_D	Measured BW	
3.10	1	1.31	18.5	15.88			
	1	0.90	14.5	17.62			
	1	0.68	13.0	20.75			
	1	0.54	10.0	22.84	12.67	12.88	
	1	0.45	8.0	24.58	14.7	13.5	
	4	1.31	17.5	9.65			
	4	0.90	15.5	13.38			
	4	0.68	14.0	17.32			
	4	0.54	12.0	21.72	12.67	12.88	
	4	0.45	11.0	25.21	14.7	13.5	
	10	1	1.21	17.0	21.32		
		1	0.90	14.0	22.55		
1		0.68	12.0	24.35			
1		0.54	10.0	26.14	12.67	12.88	
1		0.45	8.0	26.62	14.7	13.5	
4		1.31	14.5	13.80			
4		0.90	13.0	15.71			
4		0.68	12.0	17.96			
4		0.54	11.5	20.56	12.67	12.88	
4		0.45	10.0	22.82	14.7	13.5	
100		1	1.31	16.0	22.58		
		1	0.90	13.5	23.71		
	1	0.68	11.5	25.18			
	1	0.54	9.5	26.56	12.67	12.88	
	1	0.45	7.5	26.76	14.7	13.5	
	4	1.31	12.5	15.05			
	4	0.90	11.5	15.87			
	4	0.68	11.0	17.27			
	4	0.54	10.0	19.32	12.67	12.88	
	4	0.45	9.0	21.15	14.7	13.5	

$$C_1/C_2 = 0.9214$$

$$\rho_1/\rho_2 = 0.9073$$

$$\beta = 2.66^\circ$$

PROGRAM WDGBTM

WDGBTM
 PROGRAM WDGBTM USES THE METHOD OF IMAGES WITH EITHER AN INFINITELY DISTANT OR FINITE DISTANCE SOURCE TO CALCULATE THE PRESSURE AND PHASE ALONG THE BOTTOM OF A WEDGE LAYING OVER A FAST BOTTOM. THE PRESSURE AND PHASE ARE OUTPUT BEFORE IN BOTTOM CALCULATIONS ARE BEGUN. THE USED TO CALCULATE PRESSURE ALONG THE WEDGE BOTTOM ARE THEN USED TO CALCULATE PRESSURE FROM THE WEDGE BOTTOM AS A FUNCTION OF ANGLE AND RADIUS FROM THE FIRST MODE DUMPING POINT. THE OUTPUTS ARE RESPECTIVELY REFERRED TO AS THE DEPTH DOMAIN AND ANGLE DOMAIN. ALL OUTPUT IS BOTH PRINTED AND PLOTTED. ALL DIMENSIONS ARE NORMALIZED TO THE DUMP DISTANCE CAPX. ALL Q VARIABLES ARE SINGLE PRECISION VALUES TO FEED PLOT ROUTINES.

ALL CALCULATIONS FOR PRESSURE AND AMPLITUDE ALONG THE WEDGE BOTTOM ARE AN ADAPTION OF PROGRAMS BY KAWAMURA (PRESSURE ON THE INTERFACE BETWEEN A CONVERGING FLUID WEDGE AND A FAST FLUID BOTTOM, KAWAMURA, M., AND TDANNOU, M.S. THESIS, NAVAL POSTGRADUATE SCHOOL, MONTEREY, 1978).

```

IMPLICIT REAL*(A-H,D-P,R-Z),REAL*4(Q),INTEGER(I-N)
DIMENSION PA(1001),PH(1001),YY(1001),HJFXI(1001)
*   BTMPPH(1001),BTMAMP(1001),BTMPHS(1001),YNORM(1001),
*   QXLAB(3),QYLAB(2),QPLDTA(1001),QNDRY(1001),QBAMP(1001)
*   QETA(1001)
COMMON C21,RC21,BETA,WN,PI,ANGL0,F,ANGLSO
COMMON /FINE/SRCDIS,ROI,RD2,C1,C2,BETADG,Y1NORM,Y2NORM,
C12,R+012,PINC,PA,PH,YY,YVORM,DY,ANGLC,CAPX,N
COMMON /ANGLE/SIGMA,C1HBEG,C1HEND,C1INC
DATA QXLAB/4HCAPX,4HETA,4HANGL/
DATA QYLAB/4HWAMP,4HBAMP/
DATA FINI/0,FINI,0/
DATA DEPT/0,DEPT,0/
DATA ANGL/0,ANGL,0/
    
```

READ IN DATA. TYPE OF CALCULATION IS DETERMINED BY LITERAL CARDS.

100 READ(5,200,END=5000) CHECK
 200 FORMAT(4I)

0000000000000000000000000000000000

000


```

*      42X,'SOURCE ANGLE=',E15.7//
C      PI = 4.000 * DATAN(1.000)
C      CONVERT ANGLES IN DEGREES TO RADIANS.
      BETA = BETADG * PI / 180.000
      ANGL0 = ANGLS0 * PI / 180.000
      ANGLC = DARCOS(C1/C2)
      HO = C1 / (4.000 * F * DSIN(ANGLC))
      CAPX = HO / DSIN(BETA)
      WRITE(6,1060) HO,CAPX
1060 *  FORMAT(//,20X,'LOWEST POSSIBLE MODE CUT OFF DEPTH',9X,E15.7
      /20X,'DISTANCE FROM APEX ',21X,E15.7//)
C      THE RANGE WE WILL CALCULATE PRESSURE AND
      PHASE OVER IS GIVEN BY Y2 - Y1
      Y1 = Y1NORM * CAPX
      Y2 = Y2NORM * CAPX
      DY = Y2 - Y1
C      CALCULATE A SERIES OF EXPRESSIONS FOR LATER
      USE. MN IS THE WAVE NUMBER.
      WN = 2.000 * PI * F / C1
      C21 = C2 / C1
      RC21 = C2 * R02 / (C1 * R01)
      C12 = C1 / C2
      RH012 = R01 / R02
      PINC = 1.000 / DFLOAT(N)
C      WRITE LABELS FOR WEDGE BOTTOM
      WRITE(6,1070)
1070 *  FORMAT(//22X,'DISTANCE',17X,'NORMALIZED DISTANCE',6X,
      'PRESSURE AMP.',12X,'PHASE ANGLE'//)
C      CALCULATE THE PRESSURE AND PHASE AS A FUNCTION
      OF INCREMENTED DISTANCE YY.
      L = N + 1
      DO 1500 I = 1,L
      DISTY = Y1 + DFLOAT(I-1) * DY / DFLOAT(N)
      YY(I) = DISTY
      YNORM(I) = DISTY / CAPX
      CALL PRESS(DISTY,PAMP,PHAS)
      PA(I) = PAMP
      PH(I) = PHAS

```

```

1550 WRITE(6,1550) YY(I), YNORM(I), PA(I), PH(I)
1500 FORMAT( 20X,3(E15.7,10X),E15.7)
1501 CONTINUE
C
C
C      PLOT THE PRESSURE AMPLITUDE ALONG THE BOTTOM
      OF THE WEDGE.
QRH12 = RHD12
QC12 = C12
QBETA = BETAD3
QYIN = YINORM
QPINC = Y2NORM / DFLOAT(N)
NN = N + 1
C
C      SCALE AMPLITUDE FOR PLOTTING.
DQ 1505 IJK = 1,NN
QPLOTA(IJK) = PA(IJK) * 0.5D0
QNORM(IJK) = YNORM(IJK)
CONTINUE
CALL PLOIS(0,0,0)
CALL AXISMN(QNORM(1), QNORM(N+1), QXLAB(1), QYLAB(1), 0,0)
CALL CAPSHN(1)
CALL AMPLT(QNORM, QPLOTA, NN)
CALL PLOT(0,0,0,-999)
GO TO 100
C
C
C      INTEGRATION INTO THE BOTTOM
C
C      AN ATTENUATION COEFFICIENT (ALPHA) IS INCORPORATED
C      TO ANALYSE MODE INTERFERENCE EFFECTS. WHEN ALPHA
C      IS ZERO, THE CALCULATIONS ARE UNATTENUATED. THIS
C      OPTION IS NOT AVAILABLE IN ANGLE DOMAIN CALCULATIONS.
C
C
C      2000 CONTINUE
READ(5,2100) ETABEG, ETAMAX, XI, INC
2100 FORMAT(3(F15.7,5X),15)
2102 READ(5,2102) ALPHA
2150 WRITE(6,2150) ETABEG, ETAMAX, INC, XI
2150 FORMAT(//15X, 'DEPTH DOMAIN INPUT DATA', //
*          25X, 'BEGINNING ETA (Z/X) ', //10X, F5.2/
*          25X, 'MAXIMUM DEPTH ETA (Z/X) ', //10X, F5.2/
*          25X, 'DEPTH INCREMENTS USED ', //7X, 15/
*          25X, 'Y/X DISTANCE FOR PROBE ', //5X, F15.7//)
*
*
*

```

```

2103 WRITE(6,2103) ALPHA
C      FORMAT(/,20X,'ALPHA =',5X,F15.7)
C
C      NOW SET UP CONSTANT TERMS FOR LATER USE AND INITIALIZE ETA.
C
C      WN2X = PI / (2.000 * DTAN(ANGLC) * DTAN(BETA))
C      WNTRM = DSQRT(2.000 * WN2X / PI) / 4.000
C      CPI = DCOS(0.7500 * PI)
C      SPI = DSIN(0.7500 * PI)
C
C      NOW PREPARE TO INTEGRATE. THE REAL AND IMAGINARY PORTIONS
C      ARE DONE SEPARATELY.
C
C      DXI = DY / (DFLOAT(N) * CAPX)
C      ETAINC = (ETAMAX - ETABEG) / DFLOAT(INC)
C      JJ = INC + 1
C      DO 2500 KK = 1, JJ
C
C      CALCULATE THE INTEGRAL MULTIPLIERS
C
C      ETA = DFLOAT(KK-1) * ETAINC
C      PRETRM = ETA * WNTRM
C      ETASQR = ETA**2.000
C
C      NOW CALCULATE THE AMPLITUDE TERM COMMON TO BOTH THE REAL AND
C      IMAGINARY INTEGRATIONS. XI PRIME IS THE QUANTITY YY FROM THE
C      BOTTOM DOMAIN CALCULATIONS.
C
C      SUMRL = 0.000
C      SUMIM = 0.000
C      L = N + 1
C      DO 2550 LL = 1, L
C      IF(YNORM(LL).EQ.0.000-AND.ETASQR.EQ.0.000) GO TO 2540
C      XIETA = ((XI - YNORM(LL))**2 + ETASQR)
C
C      THE AMPLITUDE MAY BE ATTENUATED.
C
C      AMPTRM = PA(LL) * DEXP(-1.0 * ALPHA * DSQRT(XIETA)) /
C      * (XIETA**.75)
C
C      THE PHASE TERM IS ALSO A COMMON TERM TO BOTH INTEGRALS.
C
C      PHSTRM = PH(LL) - (WN2X * DSQRT(XIETA))
C
C      NOW SUM OVER THE REAL AND IMAGINARY PARTS.
C
C      COSPHS = DCOS(PHSTRM)
C      SINPHS = DSIN(PHSTRM)

```

```

CHIMRL = COSPHS * AMPIRM * DXI
CHIMIM = SINPTS * AMPIRM * DXI
GO TO 2545
CONTINJE
2540 CHIMRL = 0.000
CHIMIM = 0.000
CONTINJE
2545 SUMRL = SUMRL + CHIMRL
SUMIM = SUMIM + CHIMIM
CONTINJE
2550 PARTI = PRETR4 * (CPI * SUMRL - SPI * SUMIM)
PARTI = PRETR4 * (CPI * SUMRL + SPI * SUMRL)
BTMDPH(KK) = ETA
QETA(KK) = BTMDPH(KK)
BTMAMP(KK) = DSQRT(PARTI**2 + PARTI**2)
IF (PARTI.EQ.0.000) GO TO 2546
BTMPS(KK) = DATAN2(PARTI,PARTI)
GO TO 2500
2546 BTMPS(KK) = 0.000
2500 CONTINJE

C WRITE OUT THE RESULTS.
2600 WRITE(6,2600)
FORMAT(//20X,'Z DISTANCE',19X,'Z PRESSURE',12X,'PHASE ANGLE'//)
DO 2700 MM = 1,JJ
WRITE(6,2800) BTMDPH(MM),BTMAMP(MM),BTMPS(MM)
FORMAT( 20X,E15.7,10X,E15.7,10X,E15.7)
CONTINUE

C NOW PLOT THE PRESSURE AMPLITUDE INTO THE BOTTOM.
IY = 1
QETAB = ETABE3
QETAM = ETAMAX
CALL PLOT(0.,0.,1-3)
CALL AXISMR(QETAB,QETAM,QXLAB(2),QYLAB(2),0,IY)
CALL CAPSHN(2)
SCALE = DSQRT(DABS(XI - 1.000))
DO 2705 IL = 1,JJ
QBAMP(IL) = BTMAMP(IL) * 2.000 * SCALE * 5.000/6.000
CONTINJE
2705 CALL AMPPLT(QETA,QBAMP,JJI)
CALL PLOT(0.,0.,-999)
GO TO 100
CONTINUE
3000 THE FOLLOWING CODE CALCULATES THE PRESSURE AND AMPLITUDE
C IN THE BOTTOM FOR A GIVEN CONSTANT SIGMA = (R/CAPX) AND
C

```

AS A FUNCTION OF ANGLE CAPITAL THETA (THETA).

INPUT CONSISTS OF:
SIGMA = THE RADIUS NORMALIZED TO CAPX.
CTHBEG = THE BEGINNING ANGLE IN DEGREES.
CTHEND = THE ENDING ANGLE IN DEGREES.
CTHINC = THE ANGULAR INCREMENT IN DEGREES, SHOULD NOT
BE LARGER THAN 0.5 DEGREES.

READ IN INPJ

```
3100 READ(5,3100) SIGMA, CTHBEG, CTHEND, CTHINC
3110 FORMAT(4,F5.2,5X)
3150 WRITE(6,3150) SIGMA, CTHBEG, CTHEND, CTHINC
3160 FFORMAT(//15X,ANGLE,DOMAIN,INPUT DATA,//
* 25X,SIGMA (R/XI) = ,10X,F5.2/
* 25X,BEG CAP THETA = ,10X,F5.2/
* 25X,END CAP THETA = ,10X,F5.2/
* 25X,CAP THETA INC = ,10X,F5.2/)
```

NOW SET UP CONSTANT TERMS FOR LATER USE.

```
3170 RTHBEG = CTHBEG * PI / 180.000
3180 RTHEND = CTHEND * PI / 180.000
3190 RTHINC = CTHINC * PI / 180.000
3200 WN2X = PI / (2.000 * DTAN(ANGLC) * DTAN(BETA))
3210 WNTRM = DSQRT(2.000 * WN2X / PI)
3220 CPI = DCOS(0.2500 * PI)
3230 SPI = DSIN(0.2500 * PI)
```

NOW PREPARE TO INTEGRATE. THE REAL AND IMAGINARY PORTIONS
ARE DONE SEPARATELY.

```
3240 DXI = DY / (DELDTA(N) * CAPX)
3250 INC = IDINT((RTHEND - RTHBEG) / RTHINC)
3260 JJ = INC + 1
3270 DO 3500 KK = 1, JJ
```

CALCULATE THE INTEGRAL MULTIPLIERS.

```
3280 THETA = RTHBEG + ((KK - 1) * RTHINC)
3290 EPSLON = SIGMA * DCOS(THETA)
3300 TSSIGMA = DSQRT((SIGMA**2) - (EPSLON**2))
3310 PRETRM = TSSIGMA * WNTRM
3320 TSSQR = TSSIGMA**2.0
```

NOW CALCULATE THE AMPLITUDE TERM COMMON TO BOTH THE REAL
AND IMAGINARY INTEGRATIONS. XI PRIME IS THE QUANTITY YY

```
C
C FROM THE BOTTOM DOMAIN CALCULATIONS.
```

```
SUMRL = 0.000
SUMIM = 0.000
L = N + 1
DO 3550 LL = 1, L
IF(YNORM(LL).EQ.(0.000).AND.TSSQR.EQ.(0.000)) GO TO 3540
XI = 1.000 - EPSLON
XITS = ((XI - YNDRM(LL))**2 + TSSQR)
AMPTRM = PA(LL) / (XITS**0.75)
```

```
C THE PHASE TERM IS ALSO A COMMON TERM TO BOTH INTEGRALS.
```

```
C PHSTRM = PH(LL) - (WN2X * DSQRT(XITS))
```

```
C NOW SUM OVER THE REAL AND IMAGINARY PARTS.
```

```
C COSPHS = DCOS(PHSTRM)
C SINPHS = DSIN(PHSTRM)
C CHIMRL = COSPHS * AMPTRM * DXI
C CHIMIM = SINPHS * AMPTRM * DXI
GO TO 3545
```

3540

```
CHIMRL = 0.000
```

3545

```
CHIMIM = 0.000
```

```
CONTINUE
SUMRL = SUMRL + CHIMRL
SUMIM = SUMIM + CHIMIM
```

3550

```
CONTINUE
PARTI = PRETRM * (CPI * SUMRL - SPI * SUMIM)
PARTI = PRETRM * (CPI * SUMRL + SPI * SUMRL)
```

```
BTMDPH(KK) = T-ETAK * 180.000 / PI
```

```
QETA(KK) = BTMDPH(KK)
```

```
BTMAMP(KK) = DSQRT((PARTI**2) + (PARTI**2) )
```

```
IF(PARTI.EQ.0.000) GO TO 3555
```

```
BTMPS(KK) = DATAN2(PARTI, PARTI)
```

```
GO TO 3560
```

```
BTMAMP(KK) = 0.000
```

3555

```
BTMAMP(KK) = 0.000
```

```
CONTINUE
```

```
CONTINUE
```

3600

```
WRITE(6,3600)
```

```
FORMAT(/,20X,' ANGLE DOWN',19X,' R PRESSURE',12X,' PHASE ANGLE',//)
```

```
DO 3700 MM = 1, JJ
```

```
WRITE(6,3800) BTMDPH(MM), BTMAMP(MM), BTMPS(MM)
```

```
FORMAT(20X,E15.7,10X,E15.7,10X,E15.7)
```

```
CONTINUE
```

```
C NOW WRITE OUT THE RESULTS.
```

C

```

C      NOW PLOT THE AMPLITUDE AS A FUNCTION OF ANGLE
IX = 1
IY = 1
CALL PLOT(0.,0.,-3)
QBEG = CTHBEG
QEND = CTHEND
CALL AXISMR(QBEG,QEND,QXLAB(3),QYLAB(2),IX,IY)
CALL CAPSHN(3)
DO 3705 IL = 1, JJ
QBAMP(IL) = BT4AMP(IL) * 2.000 * 5.000/6.000
CONTINUE
3705 CALL APPPLT(QETA, QBAMP, JJ)
GO TO 100
CONTINUE
5000 CALL PLOTS(0.,0.,+999)
STOP
END

```

```

C C SUBROUTINE PRESS(DISTX,PAMP,PHAS)
C C   DEFINE CHARACTERISTICS OF VARIABLES, PARAMETERS, CONSTANTS
C C   IMPLICIT REAL*8(A-H,O-P,R-Z),REAL*4(Q),INTEGER(I-N)
C C   COMPLEX*16 R,RA,RB,REFL,REFLNO,P,PRES,Z
C C   COMMON C21,RC21,BETA,WN,PI,ANGLO,F,ANGLSO
C C   INITIALIZE COMPLEX PRESSURE P = 0.0 + J0.0
C C   P = (0.000,0.000)
C C   RESET COUNTER N=0, RESET IFLAG = 0.
C C   N = 0
C C   IFLAG = 0
10 CONTINUE
C C   CALCULATE ANGLN AND DETERMINE IF ANGLE IS IN THE RANGE OR NOT
C C   IF ANGLN.GE.PAI GO TO THE NEXT STEP
C C   ANGLN = 2.000 * INT((N + 1.) / 2.) * BETA + (-1)**N * ANGLO
C C   IF (ANGLN.GE.PI) GO TO 80
C C   CALCULATE THE TOTAL REFLECTION COEFFICIENT R BY EACH PATH
C C   IF(IFLAG.EQ.1) GO TO 40
C C   REFLNO = REFL(ANGLN)
C C   IF(N.GE.2) GO TO 20
C C   RA = REFLNO
C C   R = RA
C C   GO TO 30
20 CONTINUE
C C   RA = RA * REFLNO
C C   R = RA
30 CONTINUE
C C   IFLAG = 1
C C   GO TO 70
40 CONTINUE
C C   REFLNO = REFL(ANGLN)
C C   IF (N.GE.2) GO TO 50
C C   RB = REFLNO
C C   R = RB
C C   GO TO 60
50 CONTINUE
C C   RB = RB * REFLNO
C C   R = RB
60 CONTINUE

```



```

IFLAG = 0
70 CONTINUE
CC      CALCULATE PARAMETERS Z AND PRES
      Z = DCPLX(0.000,WN * DISTX * DCOS(ANGLN))
      PRES=CDEXP(Z)*(-1)**INT((N+1.)/2.)*
      * (1.000 + 1.000 / REFLN0) * R
CC      CALCULATE THE PRESSURE WHICH IS THE SUM OVER PRES.
      P = P + PRES
CC      SET TO THE NEXT PAIR OF IMAGES
      N = N + 1
      GO TO 10
80 CONTINUE
CC      CALCULATE PA AND PH, AND RETURN BACK TO THE MAIN ROUTINE.
      PAMP = CDABS(P)
      PREAL = DREAL(P)
      PIMAG = DIMAG(P)
      IF(PIMAG.EQ.0.000) GO TO 90
      PHAS = DATAN2(PIMAG,PREAL)
      GO TO 100
90 PHAS = 0.000
100 CONTINUE
      RETURN
      END

```

```

FUNCTION REFL (ANGLN)
  DEFINE CHARACTERS OF VARIABLES, PARAMETERS AND CONSTANTS
  IMPLICIT REAL*8(A-H,O-P,R-Z),REAL*4(Q),INTEGER(I-N)
  COMPLEX*16 PSAI,REFL
  COMMON C21,RC21,BETA,WN,PI,ANGLO,F,ANGLSO
  CALCULATE PARAMETER CHECK
  CHECK = 1.000 - C21**2 * DCOS(ANGLN)**2
  IDENTIFY THE GRAZING ANGLE AND CHECK TO SEE IF IT IS LESS
  THAN THE CRITICAL ANGLE.
  IF (CHECK.GT.0.000) GO TO 10
  CALCULATE THE PARAMETER PSAI FO THE REFLECTION COEFFICIENT
  OF EACH BOJNCE. THERE ARE TWO WAYS WHICH DEPEND ON THE
  IDENTIFICATION OF A CHECK.
  PSAI = DCMPLEX(0.000,-DSQRT(-CHECK)/DSIN(ANGLN))
  GO TO 20
10 CONTINUE
  PSAI = DCMPLEX(DSQRT(CHECK)/DSIN(ANGLN),0.000)
  GO TO 20
  CALCULATE REFL
  REFL = (RC21 - PSAI) / (RC21 + PSAI)
  RETURN BACK TO SUBROUTINE PRESS
RETURN
END

```

```

SUBROUTINE AXISMR(QSTR,QFIN,QXLAB,QYLAB,IX,IY)
C
C THIS SUBROUTINE DRAWS THE AXES AND WRITES THE SCALE
C MARKINGS EVERY ONE-FOURTH ETA. THE VERTICAL SCALE
C IS FIXED SO VARIOUS RUNS CAN BE OVERLAID.
C
C FIRST DRAW A BOX SIX INCHES LONG BY FIVE INCHES HIGH.
C THE BOX IS OFFSET TO ALLOW FOR LABELING.
CALL OFFSET(-1.,1.,1.,1.,1.)
CALL NEWPEN(3)
CALL PLOT(0.,0.,+13)
CALL PLOT(6.,5.,+12)
CALL PLOT(6.,5.,+12)
CALL PLOT(0.,5.,+12)
CALL PLOT(0.,5.,+12)
IF(IX.EQ.1) GO TO 150
C
C NOW LABEL THE X-AXIS EVERY ONE-FOURTH UNIT.
QSTEP = ABS( 6.0 / ((QFIN - QSTR) * 4.0) )
QMARK = QSTEP
QTICK = QSTR + 0.25
CONTINUE
CALL NEWPEN(1)
CALL PLOT(QMARK,0.,+13)
CALL PLOT(QMARK,-0.1,+12)
CALL NJMBER(QMARK+0.85,0.8,0.07,QTICK,0.0,+2)
QMARK = QMARK + QSTEP
QTICK = QTICK + 0.25
IF(QMARK.LT.6.0) GO TO 100
GO TO 170
CONTINUE
100
C
C MARK THE AXIS FOR ANGLE DOMAIN
QSTEP = ABS( 6.0 * 5.0 / (QFIN - QSTR) )
QMARK = QSTEP
QTICK = QSTR + 5.0
CONTINUE
CALL NEWPEN(1)
CALL PLOT(QMARK,0.0,+13)
CALL PLOT(QMARK,-0.1,+12)
CALL NUMBER(QMARK+0.85,0.8,0.07,QTICK,0.0,+2)
QMARK = QMARK + QSTEP
QTICK = QTICK + 5.0
IF(QMARK.LT.6.0) GO TO 160
CONTINUE
160
C
C
170

```

C C C
NOW MARK THE Y-AXIS, ONE UNIT FOR EVERY TWO INCHES
IF THIS IS FOR THE WEDGE BOTTOM.

```
IF(IY.EQ.1) G3 TO 300
QYSTEP = 1.0
QYVAL = 2.0
DO 200 I = 1,4
CALL PLOT(0.0,QYSTEP-1.0,+13)
CALL PLOT(-0.1,QYSTEP-1.0,+12)
CALL NUMBER(0.7,QYSTEP,0.07,QYVAL,0.0,+1)
QYSTEP = QYSTEP + 1.0
QYVAL = QYVAL + 2.0
200 CONTINUE
GO TO 400
300 CONTINUE
```

C C C
MARK THE AXIS FOR BOTTOM AMPLITUDE.

```
QYY = 5.0 / 6.0
QYINC = QYY
QYVAL = 0.5
DO 350 J = 1,5
CALL PLOT(-0.1,QYY,+13)
CALL PLOT(0.0,QYY,+12)
CALL NUMBER(0.5,QYY+1.0,0.07,QYVAL,0.0,+1)
QYY = QYY + QYINC
QYVAL = QYVAL + 0.5
350 CONTINUE
400 CONTINUE
```

C C C
FINALLY PRINT LABELS ON X AND Y AXES.

```
CALL SYMBOL(3.5,0.25,0.14,QXLAB,0.0,4)
CALL SYMBOL(4.5,0.25,0.14,QYLAB,0.0,-1)
CALL SYMBOL(0.3,3.0,0.14,QYLAB,90,4)
CALL SYMBOL(0.3,4.0,0.14,QYLAB,0.0,-1)
RETURN
END
```

```

SUBROUTINE CAPSHN(IITYP)
THIS SUBROUTINE PRINTS PERTINANT DATA ABOUT THE RUN, SUCH AS
SOURCE ANGLE, WEDGE ANGLE, ETC. ON THE VARIAN PLOT
IMPLICIT REAL*8(A-H,O-P,R-Z),REAL*4(Q),INTEGER(I-N)
DIMENSION PA(1001),PH(1001),YY(1001),HOFXI(1001)
* BTMDPH(1001),BTMAMP(1001),BTMPHS(1001),YNORM(1001),
* QXLAB(3),QYLAB(2),QPLOTA(1001),QNDRA(1001),QBAMP(1001)
* QETA(1001)
COMMON C21,RC21,BETA,WN,PI,ANGLO,F,ANGLSO
COMMON /FINE/SRCDIS,ROI,R02,C1,C2,BETADG,YINORM,Y2NORM,
* C12,RHO12,PINC,PA,PH,YY,YNORM,DY,ANGLC,CAPX,N
COMMON /ANGLE/SIGMA,CTHBEG,CTHEND,CTHINC
COMMON /BTM/ETABEG,ETAMAX,XI,INC
CALL PLOT(0,0,0,+13)
CALL NEWPEN(1)
CALL SYMBOL(1.25,10.0,0.07,4RRH01,0.0,4)
QRH01 = ROI
CALL NUMBER(2.5,10.0,0.07,QRH01,0.0,+4)
CALL SYMBOL(1.25,9.75,0.07,4RRH02,0.0,4)
QRH02 = R02
CALL NUMBER(2.5,9.75,0.07,QRH02,0.0,+4)
CALL SYMBOL(1.25,9.50,0.07,4HRRAT,0.0,4)
QRH12 = RHD12
CALL NUMBER(2.5,9.50,0.07,QRH12,0.0,+4)
CALL SYMBOL(1.25,9.25,0.07,4HC1,0.0,4)
QC1 = C1
CALL NUMBER(2.5,9.25,0.07,QC1,0.0,+4)
CALL SYMBOL(1.25,9.00,0.07,4HC2,0.0,4)
QC2 = C2
CALL NUMBER(2.5,8.75,0.07,QC2,0.0,+4)
CALL SYMBOL(1.25,8.50,0.07,4HCRAT,0.0,4)
QC12 = C12
CALL NUMBER(2.5,8.75,0.07,QC12,0.0,+4)
CALL SYMBOL(1.25,8.50,0.07,4HBETA,0.0,4)
QBETA = BETADG
CALL NUMBER(2.5,8.50,0.07,QBETA,0.0,+2)
CALL SYMBOL(1.25,8.25,0.07,4HSRCA,0.0,4)
QANG = ANGLSO
CALL NUMBER(2.5,8.25,0.07,QANG,0.0,+2)
CALL SYMBOL(1.25,8.00,0.07,4HSRCD,0.0,4)
QDIS = SRCDIS
IF(QDIS.EQ.0.0) QDIS = 999.
CALL NUMBER(2.5,8.00,0.07,QDIS,0.0,+2)
CALL SYMBOL(1.25,7.75,0.07,4HWINC,0.0,4)
QTNC = Y2NORM / DFLOAT(N)
CALL NUMBER(2.5,7.75,0.07,QTNC,0.0,+3)

```

C C

C

```

C
C
C
C
ITYPE = 1 IMPLIED WEDGE AMP PLOT
ITYPE = 2 IMPLIED DEPTH DOMAIN
ITYPE = 3 IMPLIED ANGLE DOMAIN

IF(ITYP.EQ.1) GO TO 1000
IF(ITYP.EQ.3) GO TO 100
CALL SYMBOL(4.0,10.0,0.07,4HEBEG,0.0,4)
QBEG = ETABEG
CALL NUMBER(5.0,10.0,0.07,QBEG,0.0,+1)
CALL SYMBOL(4.0,9.75,0.07,4HEEND,0.0,4)
QEND = ETAMAX
CALL NUMBER(5.0,9.75,0.07,QEND,0.0,+1)
CALL SYMBOL(4.0,9.50,0.07,4HEINC,0.0,4)
QINC = INC
CALL NUMBER(5.0,9.50,0.07,QINC,0.0,0)
CALL SYMBOL(4.0,9.25,0.07,4HPNT,0.0,4)
QXI = XI
CALL NUMBER(5.0,9.25,0.07,QXI,0.0,+2)
GO TO 1000

100 CONTINUE
CALL SYMBOL(4.0,10.0,0.07,4HSGMA,0.0,4)
QSIG = SIGMA
CALL NUMBER(5.0,10.0,0.07,QSIG,0.0,+2)
CALL SYMBOL(4.0,9.75,0.07,4HTBEG,0.0,4)
QBEG = CTHBEG
CALL NUMBER(5.0,9.75,0.07,QBEG,0.0,+2)
CALL SYMBOL(4.0,9.50,0.07,4HTEND,0.0,4)
QEND = CTHEND
CALL NUMBER(5.0,9.50,0.07,QEND,0.0,+2)
CALL SYMBOL(4.0,9.25,0.07,4HTINC,0.0,4)
QINC = CTHINC
CALL NUMBER(5.0,9.25,0.07,QINC,0.0,+3)
1000 RETURN
END

```

```
SUBROUTINE AMPPLT(QXAXIS,QYVAL,INCR)
DIMENSION QXAXIS(1001),QYVAL(1001)
QXSCAL = 6.0 / (QXAXIS(INCR) - QXAXIS(1))
CALL PLOT(0.,0.,+13)
CALL NEWPEN(3)
DO 100 I1 = 1, INCR
  QYVAL = (QXAXIS(I1) - QXAXIS(1)) * QXSCAL
  CALL PLOT(QXVAL,QYVAL(I1),+12)
CONTINUE
RETURN
END
```

```

SUBROUTINE FINESR
IMPLICIT REAL*8(A-H,G-P,R-Z),REAL*4(Q),INTEGER(I-N)
DIMENSION PA(1001),PH(1001),YY(1001),YNDRM(1001),HOFXI(1001)
COMMON C21,RC21,BETA,WN,PI,ANGL0,F,ANGLSO
COMMON /FINE/SRCDIS,RO1,RO2,C1,C2,BETADG,Y1NORM,Y2NORM,
* C12,RHO12,PINC,PA,PH,YY,Y1NORM,DY,AN3LC,CAPX,N
ANGLC = DARCOS (C1/C2)
HO = C1 / (4.000 * F * DSIN(ANGLC))
CAPX = HO / DSIN(BETA)
SRCDIS = SRCDIS * CAPX
C
BETA CAN BE GENERALIZED TO ANGLSO IN RADIANS
SRCDPH = SRCDIS * DSIN (BETA / 2.000)
Y1 = Y1NORM * CAPX
Y2 = Y2NORM * CAPX
WRITE(6,1000) C1,C2,RO1,RO2,BETA,BETADG,SRCDIS,
* SRCDPH,Y1NORM,Y1,Y2NORM,Y2,HO,CAPX
* FORMAT(/,25X,/,FINITE SOURCE INPUT DATA,/,
1000 * 15X,/,C1=,10X,F15.7,5X,/,C2=,10X,F15.7,5X,/,RHO1=,10X,F15.7,5X,/,RHO2=,10X,F15.7,5X,/,
* 15X,/,BETARAD=,10X,F15.7,5X,/,BETADG=,10X,F15.7,5X,/,
* 15X,/,SOURCE DIS=,10X,F15.7,5X,/,SOURCE DPH=,10X,F15.7,5X,/,
* 15X,/,Y1NORM=,10X,F15.7,5X,/,Y1=,10X,F15.7,5X,/,
* 15X,/,Y2NORM=,10X,F15.7,5X,/,Y2=,10X,F15.7,5X,/,
* 15X,/,CUT-OFF DPH=,10X,F15.7,5X,/,CAPX=,10X,F15.7,5X,/,
* DY = Y2 - Y1
PI = 4.000 * DATAN(1.000)
WN = 2.000 * PI * F / C1
C21 = C2 / C1
RC21 = C2 * RC2 / (C1 * RO1)
C12 = C1 / C2
RHO12 = RO1 / RO2
PINC = 1.000 / DFLOAT(N)
2000 * ANGL0 = BETA - DARSIN(SRCDPH/SRCDIS)
WRITE(6,2000)
* FORMAT(/,22X,/,DISTANCE,17X,/,NORMALIZED DIST,9X,
* PRESSURE AMP,12X,/,PHASE ANGLE,/)
L = N + 1
DO 3000 I = 1, L
YY(I) = Y1 + DFLOAT(I-1) * DY * PINC
DISTY = YY(I) / SRCDIS
CALL FINPR(DISTY,PAMP,PHAS,SRCDIS,SRCDPH)
PA(I) = PAMP
PH(I) = PHAS
YNDRM(I) = YY(I) / CAPX
WRITE(6,4000) YY(I),YNDRM(I),PA(I),PH(I)

```


4000 FORMAT(20X,3(F15.7,10X),F15.7)
3000 CONTINUE
RETURN
END

```

SUBROUTINE FINPR(DISTY,PAMP,PHAS,SRCDIS,SRCDPH)
IMPLICIT REAL*(A-H,O-P,R-Z),REAL*4(Q),INTEGER(I-N)
COMMON*16 PSAI,REFL,REFLNO,R,Z,PRES,P
COMMON C21,RC21,BETA,WN,PI,ANGLO,F,ANGLSO
P = (0.000,0.000)
N = 0
10 CONTINUE
  ANGLN = 2.000 * INT((N + 1.)/2.) * BETA +
  * (-1)**N*ANGLO
  IF(ANGLN-GE-PI) GO TO 50
  DD = DCOS(ANGLN) - DISTY
  THETA0 = DATAN2(DSIN(ANGLN),DD)
  DR = DSQRT(1.000 + DISTY**2 - 2.000 * DISTY * DCOS(ANGLN))
  R = (1.000,0.000)
  MM = IDINI(ANGLN/(2.000 * BETA))
  L = MM + 1
  DO 40 M = 1,L
    THETA0 = THETA0 - 2.000 * DFLOAT(M-1) * BETA
    CHECK = 1.000 - C21**2 * DCOS(THETA0)**2
    IF(CHECK.GT.0.000) GO TO 20
    PSAI = DCMPLEX(0.000,-DSQRT(-CHECK)/DSIN(THETA0))
    GO TO 30
  20 CONTINUE
  PSAI = DCMPLEX(DSQRT(CHECK)/DSIN(THETA0),0.000)
  30 CONTINUE
  REFL = (RC21 - PSAI) / (RC21 + PSAI)
  IF(M.EQ.1) REFLNO = REFL
  R = R * REFL
  40 CONTINUE
  Z = DCMPLEX(0.000,-WN * SRCDIS * DR)
  DIRET = 1.000
  PRES = DIRET * CDEXP(Z) * (-1)**INT((N + 1.)/2.) *
  * (1.000 + 1.000/REFLNO) * R / DR
  P = P + PRES
  N = N + 1
  GO TO 10
  50 CONTINUE
  PAMP = CDABS(P)
  PREAL = DREAL(P)
  PIMAG = DIMAG(P)
  PHAS = DATAN2(PIMAG,PREAL)
  RETURN
  END

```

LIST OF REFERENCES

1. Tien, P. K., and Martin, R. J., "Experiments on Light Waves in a Thin Tapered Film and a New Light-Wave Coupler", Applied Physics Letters, v. 18, p. 398-401, 1 May 71.
2. Kuznetsov, V. K., "Emergence of Normal Modes Propagating in a wedge on a Half-space from the former into the Latter", Soviet Physical Acoustics, v. 19, p. 241-245, Nov. - Dec. 1973.
3. Tien, P. K., Smolinsky, G., and Martin, R. J., "Radiation Fields of a Tapered Film and a Novel Film-to-Fiber Coupler", Transactions on Microwave Theory and Techniques, v. MTT-23, p. 79-85, January 1975.
4. University of Washington Technical Report 209, Ocean-Earth Acoustic Coupling, by Sigelman, R. A., et.al., May 1978.
5. Kawamura, M. and Ioannou, I., Pressure on the Interface Between a Converging Fluid Wedge and a Fast Fluid Bottom, M. S. Thesis, Naval Postgraduate School, Monterey, CA. 1978.
6. Coppens, A., Notes on Sound Propagation in a Fast Bottom, (Informal).
7. Butkov, E., Mathematical Physics, p. 616, Addison-Wesley Publishing Company, 1968.
8. Netzorg, G. B., Sound Transmission From a Tapered Fluid Layer into a Fast Bottom, M. S. Thesis, Naval Postgraduate School, Monterey, CA, 1977.
9. Jensen, F. B., and Kuperman, W. A., "Sound Propagation in a Wedge-Shaped Ocean with a Penetrable Bottom", Journal of the Acoustical Society of America, v. 67, p. 1564-1566, May 1980.
10. Sanders, J., The Experiment: Transmission of Acoustic Waves into a Fast Fluid Bottom from a Converging Fluid Wedge, paper presented at the Workshop on Seismic Propagation in Shallow Water, Arlington, VA, 6-7 July 1978.

INITIAL DISTRIBUTION LIST

	No. Copies
1. Defense Technical Information Center Cameron Station Alexandria, Virginia 22314	2
2. Library, Code 0142 Naval Postgraduate School Monterey, California 93940	2
3. Department Library Code 61 Department of Physics and Chemistry Naval Postgraduate School Monterey, California 93940	2
4. Department Chairman, Code 61 Department of Physics and Chemistry Naval Postgraduate School Monterey, California 93940	1
5. Mr. Robert F. Obrochta Earth and Ionospheric Program Code 464 800 N. Quincy St. Arlington, Virginia 22217	1
6. Dr. W. A. Kuperman SACLANT ASW Research Centre APO New York 09019	1
7. Dr. A. B. Coppens, Code 61Cz Department of Physics and Chemistry Naval Postgraduate School Monterey, California 93940	1
8. Dr. J. V. Sanders, 61Sd Department of Physics and Chemistry Naval Postgraduate School Monterey, California 93940	1
9. LCDR Norine A. Bradshaw 99 Corona Rd. Carmel, California 93923	1
10. Dr. David Blackstock Applied Research Laboratories The University of Texas at Austin Post Office Box 8029 Austin, Texas 78712	1