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Work was conducted under NOSC direction through contract N66001-79-C-0191 to Jaycor, Del Mar, CA, as part of the Guidelines for EMP Hardening Using Electro-Optic Techniques Program of the Defense Nuclear Agency (RAEV) under Program Element RDDA, work unit CG53. This report covers work from September 1979 to April 1980 and was approved for publication 12 June 1980.

The major Jaycor contributors to this study were WH Hardwick, WA Radasky, and TM Flanagan.

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## **METRIC CONVERSION**

To convert from

calorie per gram (cal/g)

ton (nuclear equivalent of TNT)

rad (rd)

to

gray (Gy)	10 <sup>-2</sup>
joule per kilogram(J/kg)	$4.184 \times 10^{3}$
joule (J)	4.184 × 10 <sup>9</sup>

Multiply by

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	•	Fiber optics Nuclear environment Transient radiati Electromagnetic pulse (EMP) Hardening	ion effects (TRE)
		20. ABSTRACT (Continue on reverse elde il necessary and identify by block number) Fiber optics will reduce the susceptibility of systems to a direct EMP three tradeoff between the shielding effectiveness of standard cables and that o receiver components. Long-haul ground systems require only electronics immune to em pickup and need not be buried for protection. Fiber optic to burnout and upset in systems that allow an outage time of 1 ms. In a s system vulnerability levels depend on fiber response and design margin. A not fail under the dose rates and total dose levels equivalent to a natural s second for 7 years.	eat. Cables shorter than 10m present a f shielding around critical fiber optic protection; the fiber optic cable is c susceptibility is less than that of hardwire steady-state or low-dose-rate environment, A fiber optic interface is feasible which will space environment of $(2 \times 10^{-3})$ rad(Si) per
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## SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

Fiber optic system electronics are no more vulnerable to total dose than those of hard-wired systems, which, at practical dose levels of 1 Mrad, do not appear vulnerable. In fiber optic cables, vulnerability level depends on cable type and system design margin. Insufficient data exist to estimate vulnerability levels for fiber optic systems in displacement damage-producing environments.

The report contains a list of 70 references.

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## OBJECTIVE

Determine the effectiveness of optical fiber interface techniques when applied to other system interface technologies as a hardening measure against a nuclear environment. Evaluate the effects of electromagnetic pulse (EMP) and transient radiation (TRE) on electro-optic and other electronic devices and components within military communication systems.

## RESULTS

Fiber optics will reduce the susceptibility of systems to a direct EMP threat. However, short cable lengths (less than ten metres) would present a tradeoff between the shielding effectiveness of standard cables and effectiveness of the shielding around critical components in the fiber optic receiver. For long-haul ground systems it is essential to protect only the electronics in the system since the fiber optic cable is immune to electromagnetic pick-up and need not be buried to protect it from the direct EMP threat.

For manned systems, where the associated transient radiation effects are related to personnel survivability, it is expected that a fiber optic system can easily meet the nuclear requirements. More shielding protection is necessary to eliminate any transient incapacitation of personnel than to protect the fiber optic interface. Tactical manned systems such as aircraft, mobile vans, ships, etc, are ideal candidates for fiber optic interfaces.

For systems that allow an outage time of 1 millisecond or more, caused by a prompt ionizing dose, fiber optics is less susceptible than hard-wired systems to burnout and upset. In a steady-state or low-dose-rate environment, vulnerability levels for the systems are highly dependent on the fiber response and the design margin. It is possible to design a fiber optic interface that will not have enough steady-state radiation-induced attenuation to fail when the dose rate and total dose levels are equivalent to a natural space environment,  $2 \times 10^{-3}$  rad(Si) per second for 7 years.

The electronics associated with fiber optic systems will be no more susceptible to the total dose than the electronics in a hard-wired system. However, at practical levels of total dose of one million rads(Si), no vulnerability of the cables has been detected in hardwired systems. For fiber optic cables the vulnerability level is highly dependent on the type of cable and the system design margin. At the present time, insufficient data are available to estimate the vulnerability levels for fiber optic systems in displacement damage-producing environments.

## RECOMMENDATIONS

1. Consider fiber optics in the design of any system whose cable runs are greater than 10 metres and which must adhere to military system EMP requirements.

2. Consider fiber optics in the design of any system whose recovery times are greater than 1 millisecond, and whose dose rate and total dose tolerance are within the link budget margin.

3. Consider fiber optics in the design of any system where personnel nuclear survivability requirements are the limiting factor.

4. Obtain additional data to ascertain the displacement damage-producing effects on fiber optic systems.

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# INTRODUCTION

The Department of Defense requires an ever-increasing number of systems to be hardened to radiation specifications. Many variations in the specifications exist because of different operational system applications, nuclear weapon effects, and natural (space) environments. Among the nuclear weapons effects requirements, the hardening of a system to the electromagnetic pulse (EMP) environment for most systems is very costly and timeconsuming. Therefore, the DoD has investigated, through many studies and programs, methods to reduce the cost and effort for accomplishing this task. One of the most promising methods, owing to its inherent immunity to electromagnetic effects, is the use of electrooptic techniques, primarily fiber optic coupled systems. However, these systems are vulnerable to other aspects of the nuclear environments. This report discusses the nuclear survivability tradeoffs involved when fiber optics are being considered as a hardening measure in the design of the system.

The approach taken is to calculate the radiation vulnerability levels of six classes of fiber optic systems for three different lengths of cables, thus considering a significant portion of military applications. The transient radiation effects (TRE) and EMP vulnerability levels of hard-wired technologies are compared within various radiation environments, and **advantages** and disadvantages of each technology are then identified. The final sections of the report present the data bases from which assumptions for the calculations were made, with suggestions for some of the radiation-hardening techniques that may be used for fiber optic systems.

Applications of fiber optic technology in systems that have EMP and radiation specifications are also discussed.

## COMPARISON OF NUCLEAR VULNERABILITIES IN FIBER OPTIC AND HARD-WIRED SYSTEMS

To compare the nuclear vulnerabilities of fiber optic and hard-wired data links, it is first necessary to identify those critical operating parameters that will be affected by the nuclear environment. It is then necessary to model the radiation effects on those components that determine the critical parameters. Finally, susceptibility levels for the link's operating parameters must be established.

For fiber optic systems, this process can be very complex because of the interdependence of the operating parameters on all electro-optical components. Further, the response of most components to the radiation environment is not well known or characterized, and in some cases the basic radiation effects mechanisms have not been identified. To accomplish the comparison study, it is necessary that suppositions be made as to the response of the components. For this study, the suppositions have been based either on available data for the components or on basic radiation effects mechanisms that should be applicable, as discussed under Nuclear Vulnerabilities of Electro-Optic Components. In some instances, extrapolation of the data over several orders of magnitude has been made; this is always prone to error since the radiation response may be different at widely varying radiation levels.

In the case of the hard-wired systems, more data are available on the radiation effects on the possible component technologies that may be employed in implementation of the system, a summary of which is presented in the Radiation Effects on Electronic Systems section. However, for the critical area of the response of cables to nuclear environments,

much controversy exists in the radiation effects community as to the best methods of predicting radiation response. Furthermore, cable response is highly dependent on the particular geometry of the system. For this reason, highly complex computer codes have been developed to predict cable response, none of which performs the calculations needed to work the problem in reverse order, that is, given a component vulnerability level, calculate the radiation environment which will produce that vulnerability level in the component. In this section, the coefficients for the calculation of coupling to the cables have been taken from reference 1. A range of 40 to 80 dB has been estimated for the shielding effectiveness of cables used in the systems, which should include most of the cables considered for military applications. These calculations are not intended to precisely predict the magnitude of the radiation environment, but rather to establish a lower limit of the radiation environment at which vulnerability may occur. These are linear models, and the predictions at the higher levels where nonlinear effects may occur, although possibly inaccurate, have been included to show the trends.

The following tables present calculations for various radiation environments that will cause degradation of link performance to below specified values, complete failure of the links, or upsets in the electronics. Each table is accompanied by a discussion of the estimates that have been made in the radiation response of the components. The tables present radiation component values for six hypothetical systems, four digital and two analog, for three transmission path lengths. These systems and distances were selected because they cover a wide spectrum of the applications needed by military communications systems.

## FIBER OPTIC DATA LINK RESPONSES TO RADIATION ENVIRONMENTS

The basic electro-optic system performance parameters are presented in table 1. The signal requirements are taken from the minimum signal power required to establish a signal-to-noise ratio of one, as presented in reference 2, using typical system coupling parameters. Where typical system performance specifications could be found, they were used. In digital systems, the 22-dB S/N requirement equates to an approximate bit error rate (ber) of  $10^{-9}$ . All specifications have assumed a light-emitting diode (LED) as the optical source and a pin diode as the photodetector; low-loss (<10 dB/km) fiber optic cable has been used except where noted. These specifications are not exact and are intended only to represent the values for typical signal levels for these classes of systems.

The susceptibility levels for the analog and digital systems have not been considered in the same manner. For digital systems, the upset level was selected as that level which equaled one-half the received signal. This assumes a signal midpoint detection scheme. For radiation effects that cause permanent degradation, the loss of all the design margin was used. For analog systems, the susceptibility level was defined as that level which would increase the noise or reduce the signal sufficiently to produce a 3-dB decrease in the required S/N ratio, referenced to the typical received power. For fast transient responses, the perturbation of the system recovers very quickly, and therefore, this definition of susceptibility level for analog systems may be too conservative; however, this is a customary definition for analog systems and was used in the calculations.

<sup>&</sup>lt;sup>1</sup> Vance, EF, Coupling to Shielded Cables, Wiley-Interscience, 1978

<sup>&</sup>lt;sup>2</sup> AFAL TR-45, Opto-electronics Aspects of Avionic System II, by JR Baird, May 1975

		Power Required for Optimum	Typic Oot	al Receiv ical Powe	ed.	Des	ion Maroi	Ē	Ū	scentihility 1 e	
	Rate Or	Receiver IC/N = 1)		-dB m)			(qB)		3	(M)	ž
System	Bandwidth	(m 8p-)	∧ 100 m	1 km	7 km	∧ 100 π	1 km	7 km	<100 m	1 km	7 km
Digital									•		
Pt-pt	1 Mbps	63	21	31	35 <sup>a</sup>	31	21	17	8×10°	800×10 <sup>- 9</sup>	320×10 <sup>-</sup> °
Pt-pt	10 Mbps	53	21	31	35 <sup>a</sup>	21	:	۲.	8×10 <sup>- 6</sup>	800×10 <sup>-9</sup>	320×10 <sup>-9</sup>
Pt-pt	100 Mbps	42	20	25 <sup>b</sup>	1	:	ß	I	10×10 <sup>-6</sup>	3.2×10 <sup>-6</sup>	I
Data bus	10 Mbps <sup>b</sup>	53	36	41 <sup>b</sup>	I	9	-	ł	250×10 <sup>-9</sup>	80×10 <sup>- °</sup>	I
Analog											
Video	5-10 MHz	57	21	31	35 <sup>a</sup>	14	4	0	100×10 <sup>- 9</sup>	10×10 <sup>- 9</sup>	4×10 <sup>-9</sup>
Wideband	180 MHz	37	22	27 <sup>b</sup>	ł	ŋ	0	1	1.25×10 <sup>-6</sup>	0.5×10 <sup>-6</sup>	I

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Table 1. Electro-optic system performance parameters

<sup>a</sup>Requires fiber optic cables with <2 dB/km attenuation

<sup>b</sup>Requires fiber optic cables with  $< 5 \, dB/km$  attenuation

5

To calculate the EMP vulnerability levels presented in table 2, it has been assumed that coupling and shielding loss from the incident power  $[P_I(t)]$  contained in the free-field pulse has been attenuated by 120 dB to the input of the fiber optic receiver. This level of attenuation was estimated as a level that could reasonably be obtained and maintained in the field. Of this, 60 dB is calculated for the coupling (dV/dt) from the capacitance of a typical detector package to the ground plane and 60 dB is allowed for the shielding effectiveness of the enclosure. The incident power in the EMP pulse has been defined as

$$P_{I}(t) = \frac{E(t)^2}{Z_0},$$

where

 $Z_0$  = the impedance of free space (377 ohms)

E(t) = the electrical free field.

The formula then used to calculate the magnitude of the free field is

$$|E(t)| = \left[ Z_0 P_{\text{(threat)}} \times \log^{-1} \frac{(\text{atten dB})}{10} \right]^{1/2}$$
  

$$\approx 2 \times 10^7 P_{\text{(threat)}}^{1/2} \text{ volts/meter,}$$

where  $P_{(threat)}$  has been obtained from the susceptibility levels in table 1.

For systems that have an operate-through criterion, susceptibility to the prompt radiation pulse is of interest. Table 3 presents the prompt ionization dose rate susceptibility levels of fiber optic systems, calculated with the assumption that the dose rate radiation sensitivity is determined primarily by the response of the photodetector. This is warranted because the dose rates are very low and the prompt radiation pulse width is narrow. Therefore, the total ionization doses received by the links are very small, which would result in little fiber response. The prompt dose rate at which upsets in the link will occur have been calculated from the known response of one of the least radiation-sensitive pin diodes commercially available the HP 5082-4601, manufactured by Hewlett-Packard. This diode was radiation characterized (ref 3) and has an ionization dose rate (D) induced photocurrent ( $I_p$ ) response of

$$l_{\rm D} = 5.2 \times 10^{-10} \text{ D}.$$

The vulnerability level was defined as that radiation-induced photocurrent in photodetectors that would equal the normal optical signal-induced current generated at the susceptibility level ( $P_e$ ) presented in table 1. The relationship used is

$$s = P_s \times R = I_p$$

where

**R** = the responsivity of the pin diode ( $\sim 0.5 \text{ A/W}$ ).

<sup>&</sup>lt;sup>3</sup> Hardwick, WH, and Kalma, AH, Effects of Low-Dose-Rate Radiation on Opto-Electronic Components and the Consequences on Fiber Optic Data Link Performance, IEEE Trans Nucl Sci NS-26, No 6, December 1979

		<u>, , , , , , , , , , , , , , , , , , , </u>	L	ength	
System	Rate or Bandwidth	Depth (m)	<100 m	1 km	7 km
Digital					
Pt-pt	1 Mbps	0	55.0	17.0	11.0
-		1	67.0	20.1	13.4
		2	81.8	25.3	16.4
		5	149.0	45.9	29.7
Pt-pt	10 Mbps	0	55.0	17.0	11.0
•		1	17.0	20.1	13.4
		2	81.8	25.3	16.4
		5	149.	45.9	29.7
Pt-pt	100 Mbps	0	61.0	35.0	-
•	-	1	79.4	42.7	_
		2	90.8	52.1	_
		5	165.0	94.5	-
Data bus	10 Mbps	0	10.0	5.0	-
		1	12.2	6.1	
		2	14.9	7.4	_
		5	27.0	13.5	
Analog					
Video	5-10 MHz	0	6.0	2.0	1.3
		1	7.3	2.4	1.6
		2	8.9	3.0	1.9
		5	16.2	5.4	3.5
Wideband	180 MHz	0	22.0	14.0	-
		1	26.8	17.1	_
		2	32.7	20.8	
		5	59.4	37.8	

Table 2. Electric free-field vulnerability levels (kV/m) for fiber optic data systems as a function of burial depth in the earth

**S** 5

<u> </u>		D (1		Length	<u></u>
System	Bandwidth	(m)	<100m	l km	7 km
Digital					
Pt-pt	l Mpbs	0	7.7	0.8	<sup>~</sup> 0.3
		1	$16.2 \times 10^{3}$	1.7 × 10 <sup>3</sup>	$0.6 \times 10^{3}$
		2	$34.0 \times 10^{6}$	3.5 × 10 <sup>6</sup>	1.3 × 10 <sup>6</sup>
		5	$30.4 \times 10^{15}$	$32.7 \times 10^{15}$	$12.3 \times 10^{15}$
Pt-pt	10 Mbps	0	7.7	0.8	0.3
		1	$16.2 \times 10^{3}$	$1.7 \times 10^{3}$	$0.6 \times 10^{3}$
		2	$34.0 \times 10^{6}$	$3.5 \times 10^{6}$	1.3 × 10 <sup>6</sup>
		5	$314.0 \times 10^{15}$	$32.7 \times 10^{15}$	$12.3 \times 10^{15}$
Pt-pt	100 Mbps	0	0.6	3.1	_
		1	$20.3 \times 10^{3}$	$6.5 \times 10^{3}$	_
		2	$42.3 \times 10^{6}$	13.7 × 10 <sup>6</sup>	_
		5	$392.0 \times 10^{15}$	$127.0 \times 10^{15}$	
Data bus	10 Mbps	0	0.2	0.1	-
		1	420.0	210.0	
		2	$882.0 \times 10^{3}$	$441.0 \times 10^{3}$	~
		5	$8.2 \times 10^{15}$	$4.1 \times 10^{15}$	_
Analog					
Video	5-10 MHz	0	0.1	0.01	0.003
		1	210.0	21.0	6.3
		2	$441.0 \times 10^{3}$	$44.1 \times 10^{3}$	$13.3 \times 10^{3}$
		5	$4.1 \times 10^{12}$	$408.0 \times 10^{12}$	$122.0 \times 10^{12}$
Wideband	180 MHz	0	1.3	0.1	_
		1	$2.7 \times 10^{3}$	210.	
		2	5.7 × 10 <sup>6</sup>	$441.0 \times 10^{3}$	_
	•	5	$53.1 \times 10^{15}$	$4.1 \times 10^{12}$	

Table 3. Prompt-dose-rate vulnerability levels [rd(Si)/s] at surfacefor fiber optic systems with an operate-through requirementas a function of burial depth in the earth

Therefore, the threshold ionization dose rate is

$$\dot{D} \simeq \frac{P_s R}{5.2 \times 10^{-10}} \simeq 10^9 \times P_s.$$

To determine the radiation level at which total dose accumulated at a low rate [<5 krd(Si)/s] will cause link performance degradation below the link design margin, three suppositions were made. First, the fiber response to the ionizing radiation environment has a dose rate (D) dependence. This dose rate dependence has been reported (refs 3-6) for some of the most radiation-resistant fibers that have been tested. Second, the dose rate dependence of the fiber may be characterized as similar to those in reference 3. Finally, at a given dose rate, the induced attenuation is a linear function of the accumulated total dose (D<sub>T</sub>) for total doses below 7 krd(Si). Using these assumptions and data from reference 6, the induced attenuation (A<sub>1</sub>) at room temperature may be expressed as

$$A_1(dB/m) = 3.27 \times 10^{-4} \dot{D}^{-0.56}$$
 for  $D_T < 7 \text{ krd}(Si)$  and  $\dot{D} > 5 \text{ rd}(Si)/s$ .

Using this expression and the design margins (DM) presented in table 1, the formula for calculating the total dose needed to induce an attenuation equal to the design margin is

$$D_{\rm T} = 3.06 \times 10^3 \ \dot{D}^{0.44} \times \frac{\rm DM}{\rm e}$$
.

where

 $\ell$  = the length of the fiber in metres.

For the levels presented in table 4, a dose rate of 8.3 rd(Si)/s was used; however, these levels would be valid for higher dose rates. For dose rates very much smaller than this value, the induced attenuation may not reach the design margin level, and therefore, the equation would be invalid. It should also be noted for this type of fiber that, once the radiation ceased, recovery would take place and the link would begin operating at some later time that would depend on the total dose received by the fiber and on its recovery rate.

Table 5 shows the increase in susceptibility level for total dose in a prompt pulse for these systems with several recovery times. The empirical equation developed to characterize the cable's (ITT 323) recovery from fast radiation pulses, as reported in reference 6, is

$$A_R = \ell(28 \times 10^{-3}) \log(t \times 10^5) + \ell KD_T$$
 for t>10<sup>-5</sup> s.

<sup>&</sup>lt;sup>4</sup> ERADCOM TR-28-13, Nuclear Radiation Vulnerability of Proposed Army Fiber Optics Communications System, by S Kronenberg, et al, 1978

<sup>&</sup>lt;sup>5</sup> Arimura, I, and Colwell, R, Radiation Degradation and Recovery in Long Fiber Optic Data Links, presented at 1980 Symposium on Fiber Optics in Nuclear Environment, March 25-27, 1980

<sup>&</sup>lt;sup>6</sup> Sigel, GH, Jr, et al, Radiation Response of Large-Core Polymer-Clad Silica Optical Fibers, IEEE Trans Nucl Sci NS-26, No 6, December 1979

where

 $A_{\mathbf{R}}$  = the amount of damage remaining in the cable (dB),

 $\ell$  = the length of the cable (m),

t = the time period of recovery (s),

K = a constant that predicts the initial attenuation as a function of prompt total dose.

	Rate or		Length	
System	Bandwidth	<100 m	1 km	7 km
Digital				
Pt-pt	1 Mbps	2400	160	19
Pt-pt	10 Mbps	1630	85	8
Pt-pt	100 Mbps	854	47	
Data bus	10 Mbps	466	8	-
Analog				
Video	5-10 MHz	1100	31	-
Wideband	180 MHz	388	-	

# Table 4. Low-dose-rate total-dose [rd(Si)] vulnerability levels for fiber optic systems

This cable has one of the lowest radiation-induced attenuations that have been reported. Even though it has a fairly low rate of recovery, it still has an induced attenuation that is below similar cables at the times used in the table. This formula may be combined with the performance specification to determine the total prompt dose  $(D_p)$  that may be sustained by the fiber and recover in the specified times presented in table 5. That is,

$$D_{p} = \frac{(DM + \ell) (28 \times 10^{-3}) \log (t \times 10^{5})}{K\ell} \times 10^{d/3}.$$

Here, the last term in the expression has been used to allow for the attenuation of the ionizing radiation with the burial depth (d) in the earth, which was calculated for 1MeV photons to be a factor of 10 for  $\sim 0.3$  m.

It should be noted that, for systems with longer recovery times, the susceptibility level is not strongly dependent on the design margin or on the length of the link. It is also interesting that, for any given depth, the prompt total dose vulnerability level for all the systems spans only one order of magnitude for this particular fiber optic cable.

Other environments of interest are those of natural space and steady state, such 'as that near a nuclear reactor. In these environments the opto-electronic components are subjected to constant fluxes of radiation particles (electrons, protons, gammas) that adversely affect operation of the data links (ref 3). The primary opto-electronic components that contribute to the degradation of operation of the data links are the fibers and the photo-detectors. Using the same type of fiber that was used for the analysis above and the photo-detector previously used, and assuming that the susceptibility level is that at which data

÷

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System         Bardwicht         (100         1 km         7 km         7 l00         1 km         7 km         7 l00         1 km           Digital         Bardwicht         (10)         1 km         7 km         7 km         7 l00         1 km         7 km         7 l00         1 km           Digital         Bardwicht         (10)         1 km         7 km         7 km         7 l00         1 km         7 km         7 l00         1 km           Prent         1 wbrs         0         9.7x10 <sup>10</sup> 3.5x10 <sup>10</sup> 3.5x10 <sup>10</sup> 3.3x10 <sup>10</sup> 5.3x10 <sup>10</sup>		Date Or	Canthold		רבוואנו			E			R	
Digital         Flecovery Time 1 m         Flecovery Time 1 m         Flecovery Time 1 m         Flecovery Time 1 m           Digital         Digital         0         37x10 <sup>3</sup> 2.0x10 <sup>3</sup> 15x10 <sup>3</sup> 3.8x10 <sup>3</sup> 14.0x10 <sup>3</sup> 5.3x10 <sup>3</sup> <th>System</th> <th>Bandwidth</th> <th>15 (E</th> <th><ul><li>100 m</li></ul></th> <th>r F</th> <th>7 km</th> <th>&lt; 100 m</th> <th>1 ka</th> <th>7 km</th> <th>&lt;100 m</th> <th>t F</th> <th>7 km</th>	System	Bandwidth	15 (E	<ul><li>100 m</li></ul>	r F	7 km	< 100 m	1 ka	7 km	<100 m	t F	7 km
Digital         Digital           Ryar         1 Mbbs         0         9 7×10 <sup>3</sup> 20×10 <sup>3</sup> 35×10 <sup>3</sup> 43×10 <sup>3</sup> 33×10 <sup>3</sup> 14,0×10 <sup>3</sup> 6,5×10 <sup>3</sup> Ryar         1         203×10 <sup>3</sup> 32×10 <sup>3</sup> 32×10 <sup>3</sup> 32×10 <sup>3</sup> 53×10 <sup>3</sup> 53×10 <sup>3</sup> 53×10 <sup>3</sup> 53×10 <sup>3</sup> 53×10 <sup>3</sup> 50×10 <sup>13</sup> 50×1				Rec	overy Time 1	SE	Œ	ecovery Time	1 s	Rec	overy Time 10	nin
Text         1 Moss         0         9.7×10 <sup>3</sup> 2.0×10 <sup>3</sup> 1.5×10 <sup>3</sup> 3.5×10 <sup>3</sup> 3.8×10 <sup>3</sup> 4.0×10 <sup>3</sup> 6.3×10 <sup>3</sup> 7.3×10 <sup>3</sup>	Digital											
1       20,3x10°       3,3x10°       3,3x10°       3,3x10°       3,3x10°       13,2x10°       22,3x10°       23,2x10°       24,2x10°       23,2x10°       23,2x10°       23,2x10°       23,2x10°       23,2x10°       23,2x10°       23,2x10°       23,2x10°       24,0x10°       23,2x10°       24,0x10°       24,0x10° </td <td>r'pt</td> <td>1 Mbps</td> <td>0</td> <td>9.7×10<sup>3</sup></td> <td>2.0×10<sup>3</sup></td> <td>1.5×10<sup>3</sup></td> <td>11.9×10<sup>3</sup></td> <td>4.3×10<sup>3</sup></td> <td>3.8×10<sup>3</sup></td> <td>14.0×10<sup>3</sup></td> <td>6.3×10<sup>3</sup></td> <td>5.8×10<sup>3</sup></td>	r'pt	1 Mbps	0	9.7×10 <sup>3</sup>	2.0×10 <sup>3</sup>	1.5×10 <sup>3</sup>	11.9×10 <sup>3</sup>	4.3×10 <sup>3</sup>	3.8×10 <sup>3</sup>	14.0×10 <sup>3</sup>	6.3×10 <sup>3</sup>	5.8×10 <sup>3</sup>
7       42.7x10°       9.0x10°       6.8x10°       5.25x10°       18.8x10°       16.8x10°       6.15x10°       26.0x10°       5.0x10°         7+pt       10       10       11       14.8x10°       1.5x10°       9.3x10°       13.3x10°       11.3x10°       6.0x10°       5.0x10°         7+pt       10       11       14.8x10°       1.5x10°       9.3x10°       9.3x10°       1.5x10°       19.3x10°       1.7x10°       5.00x10°       5.0x10°       5			-	20.3×10°	4.3×10 <sup>6</sup>	3.2×10°	2.5×10 <sup>6</sup>	8.9×10°	7.9×10 <sup>6</sup>	29.3×10°	13.2×10 <sup>6</sup>	12.2×10
Typer       10 Mbbss       0       7.06.10 <sup>16</sup> 60.05.10 <sup>16</sup> 60.05.10 <sup>16</sup> 500.05.10 <sup>16</sup> 260.05.10 <sup>16</sup> <			~	42.7×10 <sup>9</sup>	9.0×10°	6.8×10 <sup>9</sup>	52.5×10°	18.8×10°	16.6×10°	61.5×10°	27.8×10°	25.6×10
Type         10 Mbps         0         70x10 <sup>3</sup> 15x10 <sup>3</sup> 9.3x10 <sup>3</sup>			ഹ	400.0×10 <sup>18</sup>	80.0×10 <sup>18</sup>	60.0×10 <sup>18</sup>	500.0×10 <sup>18</sup>	200.0×10 <sup>18</sup>	200.0×10 <sup>18</sup>	570.0×10 <sup>18</sup>	260.0×10 <sup>1 8</sup>	240.0×10 <sup>1</sup>
Videband       18.100       3.7x10°       3.7x10°       3.7x10°       3.7x10°       2.57x10°       2.57x10°       2.57x10°       2.57x10°       2.57x10°       2.57x10°       2.57x10°       2.57x10°       2.57x10°       2.55x10°       2.55x		10 Mhns	c	7 0~ 10 <sup>3</sup>	1 Bv 10 <sup>3</sup>	1 5~10 <sup>3</sup>	0 3v 10 <sup>3</sup>	4 0~10 <sup>3</sup>	3 7v 10 <sup>3</sup>	11 3v10 <sup>3</sup>	6 0 × 10 <sup>3</sup>	5 8×10 <sup>3</sup>
Type       100 Mbps       0       44x10 <sup>3</sup> 73x10 <sup>5</sup> 6.6x10 <sup>3</sup> 40.0x10 <sup>18</sup> 200.0x10 <sup>18</sup> 200.0x	ž		- c	14 R×10	3 7 100	3 2 4 106	9.3X10 10.4X106	9.00 A	7.8×10	23 74106	12 7×10°	12 1×10
Type       Too Mbps       Constrols       Too Moles       Constrols       Constrols <thconstrols< th=""> <thconstrols< th=""> <t< td=""><td></td><td></td><td>- c</td><td>901-00-FC</td><td>7 01 109</td><td>0.2×10 2 2×109</td><td>901 X 1.61</td><td>17 6.109</td><td>901-01-01-01-01-01-01-01-01-01-01-01-01-0</td><td>40 04109</td><td>76 7~10<sup>9</sup></td><td>75 5410</td></t<></thconstrols<></thconstrols<>			- c	901-00-FC	7 01 109	0.2×10 2 2×109	901 X 1.61	17 6.109	901-01-01-01-01-01-01-01-01-01-01-01-01-0	40 04109	76 7~10 <sup>9</sup>	75 5410
T-pri         100 Mbps         0         4,4×10 <sup>3</sup> 1,6×10 <sup>3</sup> 6,6×10 <sup>3</sup> 3,9×10 <sup>3</sup> 5,9×10 <sup>3</sup> 5,9×10 <sup>3</sup> 5,9×10 <sup>3</sup> 5,9×10 <sup>3</sup> 5,9×10 <sup>3</sup> 5,0×10 <sup>3</sup> </td <td></td> <td></td> <td>on N</td> <td>300.0×10<sup>18</sup></td> <td>70.0×10<sup>1 %</sup></td> <td>60.0×10<sup>18</sup></td> <td>400.0×10<sup>18</sup></td> <td>200.0×10<sup>18</sup></td> <td>202.0×10<sup>18</sup></td> <td>460.0×10<sup>18</sup></td> <td>250.0×10<sup>18</sup></td> <td>240.0×10<sup>1</sup></td>			on N	300.0×10 <sup>18</sup>	70.0×10 <sup>1 %</sup>	60.0×10 <sup>18</sup>	400.0×10 <sup>18</sup>	200.0×10 <sup>18</sup>	202.0×10 <sup>18</sup>	460.0×10 <sup>18</sup>	250.0×10 <sup>18</sup>	240.0×10 <sup>1</sup>
Videband       180 MHz       0       28x10°       13x10°       240.0x10°       250.0x10°       250.0x10°       250.0x10°       250.0x10°       250.0x10°       260.0x10°	ł	100 Mbac	c	1 44103	1 6.103		e e. 103	204103		0 7~103	5 0~10 <sup>3</sup>	
Ata bus       10 Mbps       0       2.3×10°       5.4×10°       6.5×10°       2.5×10°       1.2×10°       1.5×10°       1.2×10°       <	L.P.I		<b>.</b>	01 × 1-10	01 X0.1	I	0.00.10	01.05.00	I	10 24 106	01.05.0	)
Nata bus       10 Mbps       0 $2.3 \times 10^3$ $1.5 \times 10^3$ $1.5 \times 10^3$ $1.5 \times 10^3$ $2.6 \times 10^3$ $2.6 \times 10^3$ $2.6 \times 10^3$ $2.8 \times 10^3$			- (	9.2X IU	0.4X IU	I	13.9X 10	81.UX1U	1	-01 X7.01	12.4X1U	1
Jata bus       10 Mbps       0       2.3x10 <sup>3</sup> 1.5x10 <sup>3</sup> -       30.0x10 <sup>18</sup> 200.0x10 <sup>18</sup> 24.00 <sup>4</sup> 3.7x10 <sup>3</sup> 5.8x10 <sup>3</sup> 24.00 <sup>4</sup> 5.8x10 <sup>3</sup> 24.00x10 <sup>18</sup> 200.0x10 <sup>18</sup> 270.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> 24			~	19.4×10 <sup>°</sup>	7.2×10°	I	29.1×10 <sup>°</sup>	170.0×10°	1	38.2×10 <sup>°</sup>	26.1×10 <sup>°</sup>	1
Data bus         10 Mbps         0         2.3×10 <sup>3</sup> 1.5×10 <sup>3</sup> -         4.5×10 <sup>3</sup> 3.7×10 <sup>3</sup> 5.8×10 <sup>3</sup> 2.5×10 <sup>3</sup> 1.2×1×10 <sup>6</sup> 2.20.0×10 <sup>18</sup> 5.0×10 <sup>18</sup> 2.00×10 <sup>18</sup> 2.00×10 <sup>18</sup> 2.00×10 <sup>18</sup> 2.200×10 <sup>18</sup> 2.4×10 <sup>3</sup> 2.5×10 <sup>3</sup> 2.55×10 <sup>3</sup> 2.55×10 <sup>3</sup> 2.55×10 <sup>3</sup> 2.55×10 <sup>3</sup> 2.55×10 <sup>3</sup> 2.55×10 <sup>3</sup> 2.50×10 <sup>18</sup> 2.00×10 <sup>18</sup> 2.00×10 <sup>18</sup> 2.200×10 <sup>18</sup> 2.200×10 <sup>18</sup> 2.200×10 <sup>18</sup> 2.200×10 <sup>18</sup> 2.9×10 <sup>3</sup> 2.5×10 <sup>3</sup> 2.5×			ß	200.0×10 <sup>18</sup>	70.0×10 <sup>18</sup>	١	300.0×10 <sup>18</sup>	200.0×10 <sup>18</sup>	I	350.0×10 <sup>18</sup>	240.0×10 <sup>18</sup>	ł
1       4.8x10 <sup>6</sup> 3.2x10 <sup>6</sup> -       9.4x10 <sup>6</sup> 78.0x10 <sup>6</sup> -       13.7x10 <sup>6</sup> 12.1x10 <sup>6</sup> 12.1x10 <sup>6</sup> 2       10.0x10 <sup>9</sup> 6.6x10 <sup>9</sup> -       19.8x10 <sup>9</sup> 164.0x10 <sup>9</sup> -       28.9x10 <sup>9</sup> 25.5x10 <sup>9</sup> 5       100.0x10 <sup>18</sup> 60.0x10 <sup>18</sup> -       200.0x10 <sup>18</sup> -       270.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> 6       5.10 MHz       0       5.2x10 <sup>3</sup> 1.6x10 <sup>3</sup> -       2.4x10 <sup>3</sup> -       270.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> /ideo       5.10 MHz       0       5.2x10 <sup>3</sup> 1.6x10 <sup>3</sup> -       2.4x10 <sup>3</sup> 5.9x10 <sup>3</sup> 25.8x10 <sup>3</sup> 7       1       10.9x10 <sup>6</sup> 3.3x10 <sup>6</sup> -       3.8x10 <sup>3</sup> -       240.0x10 <sup>18</sup> 7       1       10.9x10 <sup>6</sup> -       3.3x10 <sup>6</sup> -       3.8x10 <sup>3</sup> -       41.7x10 <sup>6</sup> 25.8x10 <sup>3</sup> 2       2       2       2       3.3x10 <sup>6</sup> -       3.3x10 <sup>6</sup> -       41.7x10 <sup>8</sup> 240.0x10 <sup>18</sup> 2       2       2       2       2       2       2       2       2       2       2       2       2	Jata bus	10 Mbps	0	2.3×10 <sup>3</sup>	1.5×10 <sup>3</sup>	I	4.5×10 <sup>3</sup>	3.7×10 <sup>3</sup>	I	6.5×10 <sup>3</sup>	5.8×10 <sup>3</sup>	I
2       100.x10°       6.6x10°       -       19.8x10°       164.0x10°       -       28.9x10°       25.5x10°       25.5x10°       25.5x10°       25.5x10°       25.5x10°       26.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> 25.2x10°       1.6x10°       -       28.5x10°       25.5x10°       25.5x10°       25.5x10°       25.5x10°       26.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> 24				4.8×10 <sup>6</sup>	3.2×10°	I	9.4×10°	78.0×10 <sup>6</sup>	1	13.7×10°	12.1×10°	1
Malog       5       100.0x10 <sup>18</sup> 60.0x10 <sup>18</sup> -       200.0x10 <sup>18</sup> 150.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> Malog       5-10 MHz       0       5.2x10 <sup>3</sup> 1.6x10 <sup>3</sup> -       7.4x10 <sup>3</sup> 3.8x10 <sup>3</sup> -       270.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> Nideo       5-10 MHz       0       5.2x10 <sup>3</sup> 1.6x10 <sup>3</sup> -       7.4x10 <sup>3</sup> 3.8x10 <sup>3</sup> -       9.5x10 <sup>3</sup> 5.9x10 <sup>3</sup> 2       2       2       2       2       2       2       3.3x10 <sup>6</sup> -       3.3x10 <sup>6</sup> -       9.5x10 <sup>3</sup> 5.9x10 <sup>3</sup> 5.9x10 <sup>3</sup> 2       2       2       2       2       3.3x10 <sup>6</sup> -       3.3x10 <sup>6</sup> -       3.3x10 <sup>6</sup> -       9.5x10 <sup>3</sup> 5.9x10 <sup>3</sup> 5.9x10 <sup>3</sup> 2.50.0x10 <sup>18</sup> 0.0x10 <sup>18</sup> 0.0x10 <sup>18</sup> 0.0x10 <sup>18</sup> -       3.3x10 <sup>6</sup> 1.2.3x10 <sup>6</sup> 2.5.8x10 <sup>3</sup> 2.5.8x10 <sup>4</sup>			2	10.0×10°	6.6×10°	1	19.8×10°	164.0×10°	I	28.9×10°	25.5×10°	1
Malog       5-10 MHz       0       5.2×10 <sup>3</sup> 1.6×10 <sup>3</sup> -       7.4×10 <sup>3</sup> 3.8×10 <sup>3</sup> -       9.5×10 <sup>3</sup> 5.9×10 <sup>3</sup> 7 ideo       5-10 MHz       0       5.2×10 <sup>3</sup> 1.6×10 <sup>3</sup> -       7.4×10 <sup>3</sup> 3.8×10 <sup>3</sup> -       9.5×10 <sup>3</sup> 5.9×10 <sup>3</sup> 1       10.9×10 <sup>6</sup> 3.3×10 <sup>6</sup> -       15.5×10 <sup>6</sup> 8.0×10 <sup>6</sup> -       19.9×10 <sup>6</sup> 12.3×10 <sup>6</sup> 2       22.28×10 <sup>9</sup> 7.0×10 <sup>9</sup> -       32.6×10 <sup>9</sup> 16.8×10 <sup>9</sup> -       41.7×10 <sup>9</sup> 25.8×10 <sup>9</sup> 5       202.0×10 <sup>18</sup> 60.0×10 <sup>18</sup> -       300.0×10 <sup>18</sup> 160.0×10 <sup>18</sup> -       390.0×10 <sup>18</sup> 240.0×10 <sup>18</sup> 6       2.8×10 <sup>3</sup> -       -       5.0×10 <sup>3</sup> -       390.0×10 <sup>18</sup> -       390.0×10 <sup>18</sup> 240.0×10 <sup>18</sup> 1       5.9×10 <sup>3</sup> -       -       5.0×10 <sup>3</sup> -       -       7.1×10 <sup>3</sup> -       390.0×10 <sup>18</sup> -       390.0×10 <sup>18</sup> -       390.0×10 <sup>18</sup> -       240.0×10 <sup>18</sup> -       21.4±0 <sup>9</sup> -       -       21.4±0 <sup>9</sup> -       -       21.4±0 <sup>9</sup> -       -       240.0×10 <sup>18</sup> <			ß	100.0×10 <sup>18</sup>	60.0×10 <sup>18</sup>	I	200.0×10 <sup>18</sup>	150.0×10 <sup>18</sup>	I	270.0×10 <sup>18</sup>	240.0×10 <sup>18</sup>	ł
fideo $5 \cdot 10 \text{ MHz}$ 0 $5.2 \times 10^3$ $1.6 \times 10^3$ $ 7.4 \times 10^3$ $3.8 \times 10^3$ $ 9.5 \times 10^3$ $5.9 \times 10^3$ $5.9 \times 10^3$ $5.9 \times 10^3$ $5.9 \times 10^6$ $1.3 \cdot 3.3 \times 10^6$ $ 7.4 \times 10^3$ $3.8 \times 10^3$ $ 9.5 \times 10^6$ $1.3 \cdot 3.3 \times 10^6$ $ 1.5 \cdot 5 \times 10^6$ $8.0 \times 10^6$ $ 1.3 \cdot 3.10^6$ $ 1.2 \cdot 3.10^6$ $2.2 \cdot 3.10^6$ $2.2 \cdot 3.10^3$ $2.40 \cdot 5.10^6$ $2.2 \cdot 3.10^6$ 5 $202 \cdot 5 \cdot 10^3$ $ 300 \cdot 5.10^3$ $ 300 \cdot 5.10^3$ $2.40 \cdot 5.10^3$ $2.40 \cdot 5.10^3$ $2.40 \cdot 5.10^3$ 6 $1.0 \cdot 5.1 \times 10^3$ $ 3.0 \cdot 5.1 \cdot 5.1 \times 10^3$ $ 3.1 \cdot 2.10^3$ $ 2.2 \cdot 5.1 \times 10^3$ $ 2.2 \cdot 5.1 \times 10^3$ <th< td=""><td><b>Nalog</b></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td></th<>	<b>Nalog</b>											
1       10.9x10 <sup>6</sup> 3.3x10 <sup>6</sup> -       15.5x10 <sup>6</sup> 8.0x10 <sup>6</sup> -       19.9x10 <sup>6</sup> 12.3x10 <sup>6</sup> 2       22.8x10 <sup>9</sup> 7.0x10 <sup>9</sup> -       32.6x10 <sup>9</sup> 16.8x10 <sup>9</sup> -       41.7x10 <sup>6</sup> 25.8x10 <sup>9</sup> 5       202.0x10 <sup>18</sup> 60.0x10 <sup>18</sup> -       32.6x10 <sup>9</sup> 16.8x10 <sup>9</sup> -       41.7x10 <sup>9</sup> 25.8x10 <sup>9</sup> 5       202.0x10 <sup>18</sup> 60.0x10 <sup>18</sup> -       300.0x10 <sup>18</sup> 16.8x10 <sup>9</sup> -       41.7x10 <sup>9</sup> 240.0x10 <sup>18</sup> 6       0       28x10 <sup>3</sup> -       300.0x10 <sup>18</sup> -       390.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> 7       1       5.9x10 <sup>9</sup> -       -       300.0x10 <sup>18</sup> -       -       390.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> 7       1       5.9x10 <sup>9</sup> -       -       -       390.0x10 <sup>18</sup> -       -       -       390.0x10 <sup>18</sup> -       -       390.0x10 <sup>18</sup> -       -	/ideo	5-10 MHz	0	5.2×10 <sup>3</sup>	1.6x10 <sup>3</sup>	I	$7.4 \times 10^{3}$	3.8×10 <sup>3</sup>	I	9.5×10 <sup>3</sup>	5.9×10 <sup>3</sup>	1
2 22.8×10° 7.0×10° - 32.6×10° 16.8×10° - 41.7×10° 25.8×10° 5 202.0×10 <sup>18</sup> 60.0×10 <sup>18</sup> - 300.0×10 <sup>18</sup> 160.0×10 <sup>18</sup> - 390.0×10 <sup>18</sup> 240.0×10 <sup>18</sup> Videband 180 MHz 0 2.8×10 <sup>3</sup> - 5.0×10 <sup>3</sup> - 7.1×10 <sup>6</sup> - 7.1×10 <sup>6</sup> - 7.1×10 <sup>6</sup> - 14.9×10 <sup>6</sup> - 2.0×10 <sup>18</sup> - 2.0×10 <sup></sup>			. –	10.9×10°	3.3×10°	ł	15.5×10°	8.0×10°	ł	19.9×10 <sup>6</sup>	12.3×10 <sup>6</sup>	j
5       202.0x10 <sup>18</sup> 60.0x10 <sup>18</sup> -       300.0x10 <sup>18</sup> 160.0x10 <sup>18</sup> -       390.0x10 <sup>18</sup> 240.0x10 <sup>18</sup> Videband       180 MHz       0       2.8x10 <sup>3</sup> -       -       7.1x10 <sup>3</sup> -         1       5.9x10 <sup>6</sup> -       -       10.5x10 <sup>6</sup> -       14.9x10 <sup>6</sup> -         2       12.4x10 <sup>8</sup> -       -       22.1x10 <sup>8</sup> -       -       31.2x10 <sup>6</sup> -         5       100.0x11 <sup>8</sup> -       -       22.1x10 <sup>8</sup> -       - <td></td> <td></td> <td>2</td> <td>22.8×10°</td> <td>7.0×10°</td> <td>I</td> <td>32.6×10°</td> <td>16.8×10<sup>9</sup></td> <td>1</td> <td>41.7×10°</td> <td>25.8×10°</td> <td>1</td>			2	22.8×10°	7.0×10°	I	32.6×10°	16.8×10 <sup>9</sup>	1	41.7×10°	25.8×10°	1
Wideband     180 MHz     0     2.8×10 <sup>3</sup> -     -     5.0×10 <sup>3</sup> -     -     7.1×10 <sup>3</sup> -       1     5.9×10 <sup>6</sup> -     -     10.5×10 <sup>6</sup> -     -     14.9×10 <sup>6</sup> -       2     12.4×10 <sup>9</sup> -     -     22.1×10 <sup>9</sup> -     -     31.2×10 <sup>9</sup> -       5     100.0×10 <sup>18</sup> -     -     20.0×10 <sup>18</sup> -     -     20.0×10 <sup>18</sup> -			ß	202.0×10 <sup>18</sup>	60.0×10 <sup>18</sup>	I	300.0×10 <sup>18</sup>	160.0×10 <sup>18</sup>	I	390.0×10 <sup>18</sup>	240.0×10 <sup>18</sup>	I
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Midahand	180 MH <sub>2</sub>	c	2 8~10 <sup>3</sup>	ļ	1	5 0~ 10 <sup>3</sup>	I	1	7 1v10 <sup>3</sup>	ł	1
$\begin{array}{cccccccccccccccccccccccccccccccccccc$			<b>-</b> -	5 0 10°			10 Ex106	l		14 0~106		1
			- ~	12 4×10		1 1	22 1× 10°	[ ]	1 1	31 24109	}	1
			i uo	100.0×10 <sup>18</sup>	i	I	210.0×10 <sup>18</sup>	l	I	290.0×10 <sup>18</sup>	ł	ļ

Table 5. Prompt total-dose vulnerability levels [rd(Si)] of fiber optic systems with allowances for recovery and depth of burial

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link operation will be degraded below the system specification levels, the degradation of the design margin is the sum of the fiber losses and the reduction of the S/N ratio of the link due to increased noise when these terms are expressed in decibels:

DM = 
$$\ell \times 3.27 \times 10^{-4} \dot{D}^{0.44} + 10 \log \frac{\dot{D} K_{D}^{2}}{I_{n}^{2}} + 1 (dB),$$

where

- $\ell$  = the length of the fiber (m),
- $k_D$  = the constant that relates the noise generated in the HP 5082-4601 photodetector =  $2.02 \times 10^{-9} A_{rms}/rd(Si)/s$  (ref 3),
- I<sub>n</sub> = the equivalent noise current at the input of the receiver calculated from table 6 optimum receiver power,

$$l_n = \log^{-1} \frac{(P_n \times R)}{10},$$

**R** = the responsivity of the photodetector.

The vulnerabilities for the fiber optic systems are given in table 6. It can be seen from the above equation, assuming equal dose rates on the detector and fiber, that for short cables the noise generated in the photodetector will cause link failure, whereas in longdistance links the induced attenuation in the fiber will cause the link to fail.

	Rate or		Length	_
System	Bandwidth	<100 m	1 km	7 km
Digital				
Pt-pt	1 Mbps	19	2	0.5
Pt-pt	10 Mbps	180	14	2.1
Pt-pt	100 Mbps	2200	200	
Data bus	10 Mbps	4.5	2.5	-
Analog				
Video	5-10 MHz	5.8	0.34	-
Wideband	180 MHz	3500	-	-

 Table 6. Fiber optic system dose rate vulnerability levels

 [rd(Si)/s] for steady-state environments

# HARD-WIRED DATA LINK RESPONSES TO RADIATION ENVIRONMENTS

For this portion of the report, the same six data systems have been analyzed for radiation susceptibilities. However, the frequency of operation of the data bus system has been lowered to 1 MHz. For the computations presented in the following tables, it is assumed that the links are interconnected with either twisted shielded pair (TSP) or coaxial cables that are terminated in their characteristic impedances. For those inter-

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System         Bandwich         (n)         10m         10m <th< th=""><th></th><th>Rate or</th><th>ţ</th><th></th><th>Length</th><th></th><th></th><th>Length</th><th></th></th<>		Rate or	ţ		Length			Length	
Digital         Literation          Literation         Literation	System	Bandwidth	(E)	10 m	100 m	1 km	10 m	100 m	1 km
Prior         1 Mbbs         1 25(1) <sup>2</sup> 38(10 <sup>1</sup> 16(10 <sup>1</sup> )         58(10 <sup>1</sup> 15(10 <sup>1</sup> )         58(10 <sup>1</sup> 15(10 <sup>1</sup> )         53(10 <sup>1</sup>					Burnout			Upset	
Prior         1 Mbps         0         1,2,10 <sup>1</sup> 3,8,10 <sup>1</sup> 1,8,10 <sup>1</sup> 3,8,10 <sup>1</sup> 3,8,40 <sup>1</sup> 2,8,40 <sup>1</sup> 3,8,40 <sup>1</sup> <td>CIGUA</td> <th></th> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td>	CIGUA								
Prat         1         2.8410 <sup>13</sup> 8.1x10 <sup>11</sup> 3.5x10 <sup>13</sup> 3.5x10 <sup>13</sup> 3.5x10 <sup>13</sup> 4.7x10 <sup>10</sup>	P-r	1 Mbps	0	1.2×10 <sup>9</sup>	3.8×10 <sup>8</sup>	1.6×10 <sup>9</sup> ª	1.7×10 <sup>7</sup>	5.4×10°	2.2×10 <sup>7</sup>
Prat         10 Mbbs         5 54:10 <sup>15</sup> 1,54:10 <sup>15</sup> 5,64:10 <sup>15</sup> 5,44:10 <sup>15</sup> 9,44:10 <sup>15</sup> 9,44:10 <sup>15</sup> 9,44:10 <sup>15</sup> 9,44:10 <sup>15</sup> 9,44:10 <sup>15</sup> 2,24:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 2,24:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 2,24:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 9,44:10 <sup>13</sup> 2,24:10 <sup>13</sup> 9,44:10 <sup>13</sup> 1,44:10 <sup>13</sup> 1,44			-	2.6×10 <sup>12</sup>	8.1×10 <sup>11</sup>	3.3×10 <sup>12</sup>	3.6×10 <sup>10</sup>	1.1×10 <sup>10</sup>	4.7×10 <sup>10</sup>
Prat         10 Mbsi         5         5.0x10 <sup>15</sup> 1.6x10 <sup>15</sup> 5.5x10 <sup>15</sup> 2.2x10 <sup>15</sup> 9.1x10 <sup>13</sup> 2.2x10 <sup>15</sup> 9.1x10 <sup>13</sup> 2.2x10 <sup>15</sup> 9.1x10 <sup>13</sup> 9.1x10 <sup>13</sup> 9.2x10 <sup>15</sup> 9.1x10 <sup>13</sup> 9.2x10 <sup>15</sup> 9.1x10 <sup>13</sup> 9.2x10 <sup>15</sup> 9.2x10 <sup>13</sup> 9.1x10 <sup>13</sup> 9.2x10 <sup>13</sup> 9.1x10 <sup>13</sup> 9.2x10 <sup>13</sup> <			2	5.4×10 <sup>15</sup>	1.7×10 <sup>15</sup>	7.0×10 <sup>15</sup>	7.6×10 <sup>13</sup>	2.4×10 <sup>13</sup>	9.9×10 <sup>13</sup>
Prati         10 Mbps         0         1.2×10 <sup>6</sup> 38×10 <sup>6</sup> 1.6×10 <sup>6</sup> 1.7×10 <sup>5</sup> 5.4×10 <sup>6</sup> 2.2×10 <sup>16</sup> 2.1×10 <sup>16</sup> 2.2×10 <sup>16</sup> 2.1×10 <sup>16</sup> 2.2×10 <sup>16</sup> 2.1×10 <sup>16</sup> 1.1×10 <sup>16</sup> 2.1×10 <sup>16</sup> 2.			ß	5.0×10 <sup>25</sup>	1.6×10 <sup>25</sup>	6.5×10 <sup>25</sup>	7.0×10 <sup>23</sup>	2.2×10 <sup>23</sup>	9.1×10 <sup>23</sup>
Pt-pt       100 Mbps       0       15x10 <sup>13</sup> 33x10 <sup>13</sup> 33x10 <sup>13</sup> 35x10 <sup>13</sup> 35x	Pr-of	10 Mhos	c	1 2×109	3 8×10 <sup>8</sup>	1 6×10 <sup>98</sup>	1 7×107	5 4×10 <sup>6</sup>	2 2×10 <sup>78</sup>
Pr.prt       100 Mbps       0       1,5x10 <sup>15</sup> 7,5x10 <sup>15</sup> 7,5x10 <sup>15</sup> 7,5x10 <sup>15</sup> 2,5x10 <sup>13</sup> 9,1x10 <sup>13</sup> 9,1x10 <sup>13</sup> Pr.prt       100 Mbps       0       1,6x10 <sup>15</sup> 1,5x10 <sup>15</sup> 7,5x10 <sup>15</sup> 7,5x10 <sup>15</sup> 2,5x10 <sup>13</sup> 9,1x10 <sup>13</sup> 9,1x10 <sup>13</sup> 9,1x10 <sup>13</sup> 9,1x10 <sup>13</sup> 1,1x10 <sup>13</sup> 1,1x10 <sup>15</sup> <t< td=""><td></td><th></th><td>, –</td><td>2 6x 10<sup>12</sup></td><td>8 1×10<sup>11</sup></td><td>3 3× 10<sup>12</sup></td><td>3 6×10<sup>10</sup></td><td>1 1×10<sup>10</sup></td><td>4 7×10<sup>10</sup></td></t<>			, –	2 6x 10 <sup>12</sup>	8 1×10 <sup>11</sup>	3 3× 10 <sup>12</sup>	3 6×10 <sup>10</sup>	1 1×10 <sup>10</sup>	4 7×10 <sup>10</sup>
Pt-pri         100 Mbps         0         1.6×10 <sup>13</sup> 1.6×10 <sup>13</sup> 5.5×10 <sup>13</sup> 7.5×10 <sup>13</sup> 2.5×10 <sup>13</sup> 9.1×10 <sup>13</sup> Pt-pri         100 Mbps         0         1.6×10 <sup>13</sup> 1.6×10 <sup>13</sup> 5.0×10 <sup>13</sup> 5.2×10 <sup>13</sup> 7.1×10 <sup>13</sup> 9.1×10 <sup>13</sup> Data bus         1 Mbps         0         1.6×10 <sup>13</sup> 2.2×10 <sup>13</sup> 2.2×10 <sup>13</sup> 3.1×10 <sup>13</sup> 2.2×10 <sup>13</sup> 9.1×10 <sup>13</sup> 1.2×10 <sup>13</sup> 2.2×10 <sup>13</sup> 1.2×10 <sup>13</sup> 2.2×10 <sup>13</sup> 1.2×10 <sup>13</sup> 1.2×10 <sup>13</sup> 2.2×10 <sup>13</sup> 1.2×10 <sup>13</sup> 2.2×10 <sup>13</sup> <td></td> <th></th> <td>. 6</td> <td>5.4×10<sup>15</sup></td> <td>1.7×10<sup>15</sup></td> <td>2.0×10<sup>15</sup></td> <td>2.6×10<sup>13</sup></td> <td>2.4×10<sup>13</sup></td> <td>9.9×10<sup>13</sup></td>			. 6	5.4×10 <sup>15</sup>	1.7×10 <sup>15</sup>	2.0×10 <sup>15</sup>	2.6×10 <sup>13</sup>	2.4×10 <sup>13</sup>	9.9×10 <sup>13</sup>
Prpt         100 Mbps         0         1.6x10 <sup>10</sup> 5.0x10 <sup>10</sup> 2.2x10 <sup>10</sup> 7.1x10 <sup>11</sup> 5.1x10 <sup>11</sup> 5.4x10 <sup>11</sup>			2	5.0×10 <sup>25</sup>	1.6×10 <sup>25</sup>	6.5×10 <sup>25</sup>	7.0×10 <sup>23</sup>	2.2×10 <sup>23</sup>	9.1×10 <sup>23</sup>
Totalog       1       3.3x10 <sup>13</sup> 1.0x10 <sup>13</sup> -       4.7x10 <sup>11</sup> 1.5x10 <sup>11</sup> -         Data bus       1 Mbps       0       1.2x10 <sup>6</sup> 2.0x10 <sup>26</sup> 2.0x10 <sup>26</sup> 2.9x10 <sup>14</sup> 3.1x10 <sup>14</sup> -         Data bus       1 Mbps       0       1.2x10 <sup>6</sup> 2.0x10 <sup>16</sup> 2.0x10 <sup>16</sup> 2.9x10 <sup>16</sup> 2.9x10 <sup>16</sup> -         Data bus       1 Mbps       0       1.2x10 <sup>16</sup> 3.8x10 <sup>16</sup> -       9.1x10 <sup>16</sup> 5.4x10 <sup>16</sup> -         2       5.0x10 <sup>25</sup> 1.6x10 <sup>15</sup> 1.7x10 <sup>16</sup> 3.8x10 <sup>16</sup> 1.1x10 <sup>16</sup> -       -       7.6x10 <sup>13</sup> -       -         2       5.0x10 <sup>15</sup> 1.6x10 <sup>15</sup> 1.7x10 <sup>15</sup> 3.8x10 <sup>15</sup> 2.8x10 <sup>16</sup> -       -         Video       5.10 MHz       0       1.3x10 <sup>16</sup> 1.3x10 <sup>16</sup> 8.2x10 <sup>16</sup> 1.5x10 <sup>16</sup> -       -         Video       5.10 MHz       1       2.7x10 <sup>15</sup> 3.5x10 <sup>12</sup> 3.5x10 <sup>16</sup> -       -         Video       5.3x10 <sup>16</sup> 5.3x10 <sup>16</sup> 5.3x10 <sup>16</sup> 3.5x10 <sup>16</sup> -       -         Video       5       5.3x10 <sup>16</sup> 5.3x10 <sup>16</sup> 3.	Pt-pt	100 Mbps	0	1.6×10 <sup>108</sup>	5.0×10°*	I	2.2×10 <sup>8ª</sup>	7.1×10 <sup>78</sup>	I
Z       7,0x10 <sup>16</sup> 2.2x10 <sup>16</sup> -       9,9x10 <sup>14</sup> 3,1x10 <sup>14</sup> -         Data bus       1 Mbps       0       1,2x10 <sup>6</sup> 3,8x10 <sup>16</sup> -       9,1x10 <sup>24</sup> 2,9x10 <sup>16</sup> -         Data bus       1 Mbps       0       1,2x10 <sup>6</sup> 3,8x10 <sup>16</sup> -       9,1x10 <sup>24</sup> 2,9x10 <sup>16</sup> -         Z       5       5,x10 <sup>15</sup> 1,7x10 <sup>15</sup> -       9,1x10 <sup>24</sup> 2,9x10 <sup>16</sup> -         Z       1       2,6x10 <sup>13</sup> 8,1x10 <sup>11</sup> -       3,6x10 <sup>16</sup> -       -         Z       5,4x10 <sup>15</sup> 1,7x10 <sup>15</sup> 3,1x10 <sup>16</sup> -       7,6x10 <sup>23</sup> 2,4x10 <sup>13</sup> -         Z       5,4x10 <sup>15</sup> 1,7x10 <sup>15</sup> 2,7x10 <sup>15</sup> 2,7x10 <sup>15</sup> 2,7x10 <sup>15</sup> 2,7x10 <sup>15</sup> 2,7x10 <sup>16</sup> 1,7x10 <sup>6</sup> 2,7x10 <sup>16</sup> -         Video       5       5,7x10 <sup>16</sup> 1,8x10 <sup>16</sup> 5,7x10 <sup>15</sup> 3,5x10 <sup>15</sup> 3,6x10 <sup>16</sup> -       -         Video       5       5,7x10 <sup>16</sup> 1,8x10 <sup>16</sup> 5,7x10 <sup>16</sup> 3,5x10 <sup>16</sup> -       -       -       2,0x10 <sup>16</sup> -       -       2,0x10 <sup>16</sup> -       -       2,0x10 <sup>16</sup> <				3.3×10 <sup>13</sup>	1.0×10 <sup>13</sup>	I	4.7×10 <sup>11</sup>	1.5×10 <sup>11</sup>	ł
Data bus       1 Mbps       5       6.5x10 <sup>26</sup> 2.0x10 <sup>26</sup> -       9.1x10 <sup>24</sup> 2.9x10 <sup>24</sup> -         Data bus       1 Mbps       0       1.2x10°       3.8x10 <sup>16</sup> -       9.1x10 <sup>21</sup> 5.4x10 <sup>16</sup> -         Data bus       1 Mbps       0       1.2x10°       3.8x10 <sup>16</sup> -       9.1x10 <sup>21</sup> 5.4x10 <sup>16</sup> -       -       9.1x10 <sup>20</sup> -       -			2	7.0×10 <sup>16</sup>	2.2×10 <sup>16</sup>	I	9.9×10 <sup>14</sup>	3.1×10 <sup>14</sup>	I
Data bus         1 Mbps         0         1.2x10°         3.8x10 <sup>8</sup> -         1.7x10 <sup>7</sup> 5.4x10 <sup>6</sup> -           2         5.4x10 <sup>15</sup> 1.7x10 <sup>15</sup> 2.6x10 <sup>15</sup> 1.7x10 <sup>15</sup> 2.4x10 <sup>13</sup> 2.2x10 <sup>23</sup> 2.2x10 <sup>23</sup> 3.4x10 <sup>13</sup> 2.2x10 <sup>23</sup> 3.4x10 <sup>13</sup> 2.2x10 <sup>23</sup> 3.4x10 <sup>13</sup> 2.2x10 <sup>24</sup> 1.6x10 <sup>13</sup> 2.2x10 <sup>24</sup> 2.2x10 <sup>24</sup> 1.6x10 <sup>13</sup> 2.2x10 <sup>26</sup> 3.4x10 <sup>13</sup> 2.2x10 <sup>26</sup> 3.4x10 <sup>13</sup> 2.2x10 <sup>26</sup> 3.4x10 <sup>13</sup> 2.2x10 <sup>26</sup> 3.4x10 <sup>13</sup> 2.2x10 <sup>26</sup> 1.6x10 <sup>13</sup> 2.2x10 <sup>26</sup> 1.6x10 <sup>13</sup> 2.2x10 <sup>26</sup> 2.2x10 <sup>2</sup>			ß	6.5×10 <sup>26</sup>	2.0×10 <sup>26</sup>	I	9.1×10 <sup>24</sup>	2.9×10 <sup>24</sup>	I
Maleq       1       2.6x10 <sup>12</sup> 8.1x10 <sup>11</sup> -       3.6x10 <sup>10</sup> 1.1x10 <sup>10</sup> -         2       5.4x10 <sup>15</sup> 1.7x10 <sup>15</sup> 1.7x10 <sup>15</sup> -       7.6x10 <sup>13</sup> 2.4x16 <sup>13</sup> -         2       5.0x10 <sup>25</sup> 1.6x10 <sup>25</sup> 1.6x10 <sup>25</sup> -       7.6x10 <sup>13</sup> 2.4x16 <sup>13</sup> -         Analog       5       5.0x10 <sup>25</sup> 1.6x10 <sup>25</sup> 1.5x10 <sup>16</sup> 1.1x10 <sup>9</sup> 2.4x16 <sup>13</sup> -         Video       5.10 MHz       0       1.3x10 <sup>16</sup> 4.1x10 <sup>9</sup> 1.3x10 <sup>9</sup> 8.2x10 <sup>45</sup> 8.2x10 <sup>45</sup> -         7/st01       1       2.7x10 <sup>13</sup> 8.6x10 <sup>12</sup> 2.7x10 <sup>13</sup> 3.6x10 <sup>11</sup> 7.2x10 <sup>36</sup> 2       5.3x10 <sup>26</sup> 1.8x10 <sup>16</sup> 5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 3.3x10 <sup>21</sup> 6.7x10 <sup>16</sup> 2       5.3x10 <sup>26</sup> 1.8x10 <sup>16</sup> 5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 3.6x10 <sup>11</sup> 7.2x10 <sup>10</sup> 2       5.3x10 <sup>16</sup> 1.5x10 <sup>16</sup> 5.3x10 <sup>16</sup> 5.3x10 <sup>16</sup> 5.3x10 <sup>16</sup> 7.1x10 <sup>7</sup> 5.2x10 <sup>16</sup> 2       5.3x10 <sup>16</sup> 2.2x10 <sup>16</sup> 2.2x10 <sup>16</sup> 2.1x10 <sup>16</sup> 2.1x10 <sup>16</sup> 1.5x10 <sup>16</sup> 1.	Data bus	1 Mbps	0	1.2×10°	3.8×10 <sup>5</sup>	I	1.7×10 <sup>7</sup>	5.4×10 <sup>6</sup>	I
Analog       5       5.4x10 <sup>15</sup> 1.7x10 <sup>15</sup> -       7.6x10 <sup>13</sup> 2.4x16 <sup>13</sup> -         Analog       5       5.0x10 <sup>25</sup> 1.6x10 <sup>25</sup> 1.6x10 <sup>25</sup> 2.2x10 <sup>23</sup> 2.2x10 <sup>23</sup> -         Analog       1       3.1x10 <sup>10b</sup> 4.1x10 <sup>9b</sup> 1.3x10 <sup>10b</sup> 4.1x10 <sup>9b</sup> 8.2x10 <sup>9b</sup> 8.2x10 <sup>4b</sup> 1.6x10 <sup>25</sup> -         Video       5.10 MHz       0       1.3x10 <sup>10b</sup> 4.1x10 <sup>9b</sup> 1.3x10 <sup>12</sup> 1.7x10 <sup>9</sup> 1.7x10 <sup>9</sup> 3.4x10 <sup>1</sup> 2       5.3x10 <sup>26</sup> 1.3x10 <sup>16</sup> 1.8x10 <sup>16</sup> 5.7x10 <sup>12</sup> 3.5x10 <sup>21</sup> 3.5x10 <sup>11</sup> 7.2x10 <sup>10</sup> 2       5.3x10 <sup>26</sup> 1.7x10 <sup>26</sup> 5.3x10 <sup>25</sup> 3.3x10 <sup>22</sup> 3.3x10 <sup>21</sup> 6.7x10 <sup>10</sup> 2       5.3x10 <sup>16</sup> 1.0x10 <sup>16</sup> 5.3x10 <sup>16</sup> 5.3x10 <sup>16</sup> 7.1x10 <sup>78</sup> -         2       5.3x10 <sup>16</sup> 1.0x10 <sup>16</sup> 2.2x10 <sup>16</sup> 5.3x10 <sup>26</sup> 3.3x10 <sup>21</sup> 6.7x10 <sup>10</sup> 2       5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 6.7x10 <sup>10</sup> -         8       5       5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 5.3x10 <sup>26</sup> 3.3x10 <sup>21</sup> 0.1x10 <sup>16</sup> <td></td> <th></th> <td>-</td> <td>2.6×10<sup>12</sup></td> <td>8.1×10<sup>11</sup></td> <td>1</td> <td>3.6×10<sup>10</sup></td> <td>1.1×10<sup>10</sup></td> <td>ı</td>			-	2.6×10 <sup>12</sup>	8.1×10 <sup>11</sup>	1	3.6×10 <sup>10</sup>	1.1×10 <sup>10</sup>	ı
Analog       5.0x10 <sup>25</sup> 1.6x10 <sup>25</sup> 1.6x10 <sup>25</sup> 1.6x10 <sup>25</sup> 2.2x10 <sup>23</sup> 2.2x10 <sup>23</sup> -         Analog       5.10 MHz       0       1.3x10 <sup>9b</sup> 4.1x10 <sup>9b</sup> 1.3x10 <sup>9b</sup> 8.2x10 <sup>4b</sup> 1.6x10 <sup>4b</sup> 3.4x10 <sup>7</sup> -         Video       5.10 MHz       0       1.3x10 <sup>9b</sup> 4.1x10 <sup>9b</sup> 1.3x10 <sup>9b</sup> 8.2x10 <sup>4b</sup> 1.6x10 <sup>4b</sup> 3.4x10 <sup>7</sup> 2       7.00 MHz       0       1.3x10 <sup>9b</sup> 1.3x10 <sup>9b</sup> 8.2x10 <sup>4b</sup> 8.2x10 <sup>4b</sup> 3.4x10 <sup>7</sup> 2       5.3x10 <sup>1a</sup> 8.6x10 <sup>1a</sup> 2.7x10 <sup>1a</sup> 2.7x10 <sup>12</sup> 1.7x10 <sup>8</sup> 3.4x10 <sup>7</sup> 2       5.3x10 <sup>2b</sup> 1.7x10 <sup>2b</sup> 5.3x10 <sup>2b</sup> 3.3x10 <sup>12</sup> 3.3x10 <sup>1a</sup> 6.7x10 <sup>1b</sup> 2       5.3x10 <sup>2b</sup> 1.0x10 <sup>1a</sup> 5.0x10 <sup>2b</sup> 5.3x10 <sup>2b</sup> 5.3x10 <sup>2b</sup> 3.3x10 <sup>2b</sup> 3.3x10 <sup>1a</sup> 6.7x10 <sup>1b</sup> Wideband       180 MHz       0       1.6x10 <sup>1a</sup> 2.2x10 <sup>1b</sup> -       2.2x10 <sup>1b</sup> -       2.2x10 <sup>1b</sup> -       2.2x10 <sup>1b</sup> -       -       2.2x10 <sup>1b</sup> -       -       -       2.2x10 <sup>1b</sup> -       -       -       2.2x10 <sup>1b</sup> -       - </td <td></td> <th></th> <td>2</td> <td>5.4×10<sup>15</sup></td> <td>1.7×10<sup>15</sup></td> <td>I</td> <td>7.6×10<sup>13</sup></td> <td>2.4×10<sup>13</sup></td> <td>I</td>			2	5.4×10 <sup>15</sup>	1.7×10 <sup>15</sup>	I	7.6×10 <sup>13</sup>	2.4×10 <sup>13</sup>	I
Analog         S-10 MHz         0         1.3x10 <sup>10</sup> 4.1x10 <sup>9</sup> 1.3x10 <sup>9</sup> 8.2x10 <sup>46</sup> 1.6x10 <sup>46</sup> 1.6x10 <sup>46</sup> Video         5-10 MHz         0         1.3x10 <sup>10</sup> 4.1x10 <sup>9</sup> 1.3x10 <sup>91</sup> 8.2x10 <sup>46</sup> 1.6x10 <sup>46</sup> 3.4x10 <sup>7</sup> 2         5.7x10 <sup>13</sup> 8.6x10 <sup>12</sup> 2.7x10 <sup>15</sup> 3.5x10 <sup>15</sup> 3.5x10 <sup>16</sup> 1.7x10 <sup>9</sup> 3.7x10 <sup>16</sup> 7.2x10 <sup>10</sup> 7.1x10 <sup>70</sup> 6.7x10 <sup>10</sup> 7.1x10 <sup>70</sup> 1.5x10 <sup>10</sup> 7.1x10 <sup>70</sup> 1.5x10 <sup>10</sup> <td></td> <th></th> <td>ß</td> <td>5.0×10<sup>25</sup></td> <td>1.6×10<sup>25</sup></td> <td>I</td> <td>7.0×10<sup>23</sup></td> <td>2.2×10<sup>23</sup></td> <td>I</td>			ß	5.0×10 <sup>25</sup>	1.6×10 <sup>25</sup>	I	7.0×10 <sup>23</sup>	2.2×10 <sup>23</sup>	I
Video         5-10 MHz         0 $1.3 \times 10^{10}$ $4.1 \times 10^{9}$ $1.3 \times 10^{9}$ $8.2 \times 10^{16}$ $1.6 \times 10^{10}$ $3.4 \times 10^{7}$ 1         2         7 \times 10^{13}         8.6 \times 10^{12}         2.7 \times 10^{9}         1.7 \times 10^{9}         1.7 \times 10^{9}         1.7 \times 10^{9}         1.7 \times 10^{9}         3.4 \times 10^{7} $3.4 \times 10^{7}$ $3.4 \times 10^{16}$ $7.2 \times 10^{10}$ $7.1 \times 10^{26}$ $7.1 \times 10^{2$	Analog								
1 $2.7 \times 10^{13}$ $8.6 \times 10^{12}$ $2.7 \times 10^{12}$ $1.7 \times 10^{9}$ $1.7 \times 10^{8}$ $3.4 \times 10^{7}$ 2 $5.7 \times 10^{16}$ $1.8 \times 10^{16}$ $5.7 \times 10^{15}$ $3.5 \times 10^{12}$ $3.6 \times 10^{11}$ $7.2 \times 10^{10}$ 2 $5.3 \times 10^{26}$ $5.3 \times 10^{26}$ $5.3 \times 10^{25}$ $3.5 \times 10^{12}$ $3.6 \times 10^{11}$ $7.2 \times 10^{10}$ 5 $5.3 \times 10^{26}$ $1.7 \times 10^{26}$ $5.3 \times 10^{25}$ $3.3 \times 10^{21}$ $6.7 \times 10^{20}$ Wideband       180 MHz       0 $1.6 \times 10^{10}$ $5.0 \times 10^{26}$ $5.3 \times 10^{25}$ $3.3 \times 10^{21}$ $6.7 \times 10^{20}$ 2 $3.3 \times 10^{13}$ $1.0 \times 10^{13}$ $1.0 \times 10^{13}$ $ 4.7 \times 10^{11}$ $1.5 \times 10^{14}$ $-$ 2 $7.0 \times 10^{16}$ $2.2 \times 10^{16}$ $ 9.9 \times 10^{14}$ $ -$ 5 $6.5 \times 10^{26}$ $2.0 \times 10^{26}$ $ 9.1 \times 10^{24}$ $ -$ 10 $6.5 \times 10^{26}$ $ 9.1 \times 10^{24}$ $2.9 \times 10^{24}$ $ -$	Video	5-10 MHz	0	1.3x10 <sup>10b</sup>	4.1×10 <sup>9b</sup>	1.3x10 <sup>9b</sup>	8.2×10 <sup>5 b</sup>	8.2×10 <sup>4b</sup>	1.6×10 <sup>4b</sup>
2       5.7×10 <sup>16</sup> 5.7×10 <sup>15</sup> 3.5×10 <sup>12</sup> 3.6×10 <sup>11</sup> 7.2×10 <sup>10</sup> 5       5.3×10 <sup>26</sup> 1.7×10 <sup>26</sup> 5.3×10 <sup>25</sup> 3.3×10 <sup>22</sup> 3.6×10 <sup>21</sup> 6.7×10 <sup>20</sup> 6.7×10 <sup>26</sup> 5.3×10 <sup>26</sup> 5.3×10 <sup>26</sup> 5.3×10 <sup>25</sup> 3.3×10 <sup>21</sup> 6.7×10 <sup>20</sup> 6.7×10 <sup>26</sup> 5.3×10 <sup>26</sup> 5.3×10 <sup>26</sup> 5.3×10 <sup>26</sup> 5.3×10 <sup>26</sup> 3.3×10 <sup>21</sup> 6.7×10 <sup>20</sup> 7       1       3.3×10 <sup>13</sup> 1.0×10 <sup>13</sup> -       2.2×10 <sup>16</sup> -       4.7×10 <sup>11</sup> 1.5×10 <sup>11</sup> -         2       7.0×10 <sup>16</sup> 2.2×10 <sup>16</sup> -       9.9×10 <sup>14</sup> 3.1×10 <sup>14</sup> -       -         5       6.5×10 <sup>26</sup> 2.0×10 <sup>26</sup> -       9.1×10 <sup>24</sup> 2.9×10 <sup>24</sup> -       -				2.7×10 <sup>13</sup>	8.6×10 <sup>12</sup>	2.7×10 <sup>12</sup>	1.7×10°	1.7×10 <sup>8</sup>	3.4×10 <sup>7</sup>
5     5.3×10 <sup>26</sup> 1.7×10 <sup>26</sup> 5.3×10 <sup>25</sup> 3.3×10 <sup>21</sup> 6.7×10 <sup>20</sup> Wideband     180 MHz     0     1.6×10 <sup>108</sup> 5.0×10 <sup>98</sup> 7.1×10 <sup>78</sup> -       2     7.0×10 <sup>113</sup> 1.0×10 <sup>13</sup> -     4.7×10 <sup>11</sup> 1.5×10 <sup>14</sup> -       2     7.0×10 <sup>16</sup> 2.2×10 <sup>16</sup> -     9.9×10 <sup>14</sup> -     -       5     6.5×10 <sup>26</sup> 2.0×10 <sup>26</sup> -     9.1×10 <sup>24</sup> 2.9×10 <sup>24</sup> -			2	5.7×10 <sup>16</sup>	1.8×10 <sup>16</sup>	5.7×10 <sup>15</sup>	3.6×10 <sup>12</sup>	3.6×10 <sup>11</sup>	7.2×10 <sup>10</sup>
Wideband         180 MHz         0         1.6x10 <sup>106</sup> 5.0x10 <sup>76</sup> -         2.2x10 <sup>86</sup> 7.1x10 <sup>76</sup> -           1         3.3x10 <sup>13</sup> 1.0x10 <sup>13</sup> -         4.7x10 <sup>11</sup> 1.5x10 <sup>11</sup> -           2         7.0x10 <sup>16</sup> 2.2x10 <sup>16</sup> -         9.9x10 <sup>14</sup> 3.1x10 <sup>14</sup> -           5         6.5x10 <sup>26</sup> 2.0x10 <sup>26</sup> -         9.1x10 <sup>24</sup> 2.9x10 <sup>14</sup> -			ß	5.3×10 <sup>26</sup>	1.7×10 <sup>26</sup>	5.3×10 <sup>25</sup>	3.3×10 <sup>22</sup>	3.3×10 <sup>21</sup>	6.7×10 <sup>20</sup>
1 3.3x10 <sup>13</sup> 1.0x10 <sup>13</sup> – 4.7x10 <sup>11</sup> 1.5x10 <sup>11</sup> – 2 7.0x10 <sup>16</sup> 2.2x10 <sup>16</sup> – 9.9x10 <sup>14</sup> 3.1x10 <sup>14</sup> – 5 6.5x10 <sup>26</sup> 2.0x10 <sup>26</sup> – 9.1x10 <sup>24</sup> 2.9x10 <sup>24</sup> –	Wideband	180 MHz	0	1.6x10 <sup>108</sup>	5.0×10 <sup>9ª</sup>	1	2.2×10 <sup>8</sup>	7.1×10 <sup>7®</sup>	1
2 7.0x10 <sup>16</sup> 2.2x10 <sup>16</sup> – 9.9x10 <sup>14</sup> 3.1x10 <sup>14</sup> – 5 6.5x10 <sup>26</sup> 2.0x10 <sup>26</sup> – 9.1x10 <sup>24</sup> 2.9x10 <sup>24</sup> –			•	3.3×10 <sup>13</sup>	1.0×10 <sup>13</sup>	I	4.7×10 <sup>11</sup>	1.5×10 <sup>11</sup>	1
5 6.5x10 <sup>26</sup> 2.0x10 <sup>26</sup> – 9.1x10 <sup>24</sup> 2.9x10 <sup>24</sup> –			2	7.0x10 <sup>16</sup>	2.2×10 <sup>16</sup>	I	9.9×10 <sup>14</sup>	3.1×10 <sup>14</sup>	ł
			5	6.5×10 <sup>26</sup>	2.0×10 <sup>26</sup>	1	9.1×10 <sup>24</sup>	2.9×10 <sup>24</sup>	1

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Table 7. Dose rate vulnerability levels [rd(Si)/s] for buried hard-wired cables

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<sup>a</sup>50-ohm coaxial systems <sup>b</sup>75-ohm coaxial systems

	Rate Or	Denth		Length			Length	
System	Bandwidth	(E)	10 m	100 m	1 km	10 E	100 m	1 ka
Digital								
Pt-pt	1 Mbp	0	9.1	0.04	0.06 <sup>b</sup>	0.13	0.5×10 <sup>-3</sup>	0.9×10 <sup>-3b</sup>
			$(910)^{a}$	(3.7)	(6.3)	(13)	(0.05)	(60.0)
		-	11.1	4.5	<i>T.T</i>	0.16	0.06	0.11
			(1110)	(450)	(770)	(15.9)	(6.1)	(11)
		2	14.8	5.5	9.4	0.19	0.07	0.13
			(1480)	(550)	(940)	(19.3)	(7.4)	(13.4)
		5	24.5	<b>6</b>	17.0	0.35	0.13	0.24
			(2450)	(1000)	(1700)	(35.1)	(13.5)	(24.3)
Pt-pt	10 Mbps	0	9.1	0.06	0.06 <sup>b</sup>	0.13	9×10 <sup>-3</sup>	0.9×10 <sup>-3b</sup>
			(010)	(6.3)	(6.3)	(13)	(60.0)	(60.0)
		-	1.11	7.7	7.7	0.16	0.11	0.11
			(1110)	(170)	()20)	(15.9)	(11)	(11)
		2	14.8	9.4	9.4	0.19	0.13	0.13
			(1480)	(040)	(940)	(19.3)	(13.4)	(13.4)
		2	24.5	17.0	17.0	0.35	0.24	0.24
			(2450)	(1700)	(1700)	(35.1)	(24.3)	(24.3)
Pt-pt	100 Mbps	0	15.8 <sup>b</sup>	0.06 <sup>b</sup>	ł	0.22	9×10 <sup>-3</sup>	ì
			(1580)	(6.3)	I	(22.5)	(60.0)	I
		-	19.3	7.7	I	0.27	0.11	I
			(1930)	(077)	ł	(27.4)	(11)	I
		2	23.5	9.4	I	0.33	0.13	ł
			(2350)	(040)	I	(33.5)	(13.4)	I
		ß	42.7	17.0	1	0.61	0.24	I
			(4270)	(1700)	1	(60.8)	(24.3)	I
Data bus	10 Mbps	0	9.1	0.04	0.06 <sup>b</sup>	0.13	0.5×10 <sup>-3</sup>	0.9×10 <sup>-3b</sup>
			(010)	(3.7)	(6.3)	(13)	(0.05)	(0.09)
		-	11.1	4.5	7.7	0.16	0.06	0.11
			(1110)	(450)	(022)	(15.9)	(6.1)	(11)
		2	14.8	5.5	9.4	0.19	0.07	0.13
		I	(1480)	(220)	(940)	(19.3)	(7.4)	(13.4)
		'n	24.5	10	17.0	0.35	0.13	0.24
			(2450)	(1000)	(1700)	(35.1)	(13.5)	(24.3)

Table 8. Hard-wired cable EM P vulnerability levels (kV/m) as a function of depth of burial

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Table 8 (cont)

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		Rate or	Darth		Length			Length	
	System	Bandwidth	(L)	10 a	100 m	1 km	10 m	100 m	1 km
	Analog								
	Video	5-10 MHz	0	13 <sup>c</sup>	0.05 <sup>c</sup>	0.05 <sup>c</sup>	0.18 <sup>c</sup>	7×10 <sup>-3c</sup>	5×10 <sup>-3c</sup>
				(1300)	(5.2)	(5.2)	(18)	(0.07)	(0.05)
			-	15.9	6.1	6.1	0.22	0.08	0.06
				(1590)	(610)	(610)	(22)	(8.5)	(6.1)
			2	19.3	74.4	74.4	0.27	0.1	0.07
				(1930)	(740)	(740)	(26.8)	(10.4)	(7.4)
			2	35.1	135	135	0.49	0.19	0.13
				(3510)	(1350)	(1350)	(48.6)	(18.9)	(13.5)
	Wideband	180 MHz	0	15.8 <sup>b</sup>	0.06 <sup>b</sup>	I	0.22	9×10 <sup>-3</sup>	I
				(1580)	(6.3)	,	(22.5)	(0.09)	1
			-	19.3	7.7	ł	0.27	0.11	I
				(1930)	(170)	J	(27.4)	(11)	I
1			2	23.5	9.4	ļ	0.33	0.13	1
15				(2350)	(040)	I	(33.5)	(13.4)	ł
			2	42.7	17.0	ł	0.61	0.24	I
				(4270)	(1700)	I	(60.8)	(24.3)	I
	Multiplier for cable 10 m above ground			0.1	01	10	01	10	10
	(referenced to $d = 0$ )				2	?	į	2	2

 $^{a}($  ) for cables with 80 dB  $S_{E};$  all others 40 dB  $S_{E}$ 

<sup>b</sup>50-ohm coaxial systems

<sup>C</sup>75-ohm coaxial systems

connected with TSP, the termination value of 150 ohms was used. The coaxial systems were assumed to be 50-ohm terminated except for the 5 to 10-MHz video system, which used 75 ohms for its termination. Further, no allowances were made for protection filters or surge suppression devices at the interface connections between the cables and the semiconductors. In the tables, the systems that use coaxial cables are identified with an asterisk. Cables were selected by considering the transmission rates or bandwidths and the lengths of cables.

Specific types of semiconductors that could be used in the implementation of the data links have not been considered for this report. It was assumed that standard IC technologies would be incorporated for both the digital and analog systems. It should be noted, however, that the 100-Mbps point-to-point digital link would probably have to be implemented with an IC technology similar to emitter-coupled logic (ECL, MECL, etc). The 180-MHz wideband link would probably be implemented using hybrid IC or discrete transistors. This is emphasized because the minimum burnout threshold for all the systems has been set at an energy level of  $5 \times 10^{-6}$  J. The range for the burnout threshold for specific devices covers approximately two orders of magnitude above this value, but it was decided that using this minimum-threshold value would result in a minimum susceptibility level and was, therefore, warranted in this study.

The upset levels for the analog and digital systems were again considered separately. For the digital systems, the upset energy threshold was set at  $1 \times 10^{-9}$  J. Interested readers who are unfamiliar with the concept of using energy to define both the burnout and upset levels, see reference 7 for a brief introductory discussion.

The upset levels for the analog systems have been set in the same manner as those in the fiber optic receivers. In fact, the same susceptibility levels were used in the calculations because the electrical power requirements for the hard-wired systems should be the same as those for fiber optic systems.

To calculate the ionization dose rates (table 7) that would create system-generated EMP (SGEMP) cable signals large enough to exceed the vulnerability thresholds of the semiconductors, the following assumptions were made. The energy  $(W_I)$  incident at the terminations of the cables could not exceed the threshold values. The energy at these points was defined as

$$W_{I}(t) = \int_{0}^{t} p(t) dt = \int_{0}^{t} e(t) i(t) dt = Z_{0} \int_{0}^{t} i^{2}(t) dt$$
  

$$\approx Z_{0} I^{2} t_{(eff)}.$$

where

p(t), e(t), and i(t) = the power, voltage, and currents, respectively, generated at the terminations as a function of time,

 $Z_0$  = the cable termination impedance,

 $t_{(eff)}$  = the electrical length of the cable.

<sup>&</sup>lt;sup>7</sup> NOSC Report TR-469, EMP Hardening of Airborne Systems through Electro-optical Techniques: Design Guideline, by R Greenwell, 15 December 1979

The current function (1) used in this equation is the magnitude of current that would be generated in an effective length  $[l_{(eff)}]$  of the cable. The effective length of cable was defined as the velocity of propagation  $(v_{prop})$  in the cable divided into the pulse width of the prompt pulse  $(t_{pw})$ . The resulting formula for the magnitude of the current is then

 $I = K\dot{D}\ell_{(eff)} = K\dot{D} t_{pw} \times v_{prop},$ 

where

K = the generation constant for the cable (C/rd-cm),

 $\dot{D}$  = the dose rate [rd(material)/s].

If the pulse width is expressed in nanoseconds and the velocity of propagation is  $\sim 0.25$  m/ns, then

$$I = \frac{K\dot{D}}{4} t_{pw}$$

It has been assumed that this current will flow for a period equal to the electrical length of the cable, and, therefore, the energy may be written as

$$W_{I} = I^{2}Z t_{(eff)} = K^{2}\dot{D}^{2} \ell^{2}_{(eff)} Z t_{(eff)}$$
$$= \frac{K^{2}\dot{D}^{2} t_{pw}^{2} \ell Z}{4}.$$

Solving the last equation for the dose rate yields

$$\dot{\mathbf{D}} = \frac{6.32 \times 10^3}{\mathrm{Kt_{pw}}} \sqrt{\frac{\mathrm{W_I}}{\mathrm{gz}}},$$

where

t<sub>pw</sub> = nanoseconds ℓ = metres.

The attenuation  $(A_{\gamma})$  of the radiation pulse through the soil may be taken into account by multiplying this expression by

$$A_{\gamma} \simeq 10^{d/0.3} \qquad (1 \,\mathrm{MeV}, \gamma),$$

where

d = the depth of the soil in metres.

For the following, estimates of peak cable-shield currents induced by EMP are used as the starting point for various coupling calculations. This section describes the bases for those estimates.

A cable of  $9.5 \times 10^{-3}$ -m diameter and lengths of 10, 100, and 1000m was assumed. The cable was either 10 m above the ground, very near the ground, or 1, 2, and 5 m below the ground surface. The ground conductivity was assumed to have a typical value of  $10^{-2}$  mho/m. The driving EMP signal was the commonly used double exponential time history with roughly a 10-ns rise time and a 250-ns decay time. The direction of EMP incidence was chosen to give maximum cable currents.

In most cases the formulas used to estimate cable currents were taken from reference 1. (This book presents a good review of EMP coupling to long cables.) Infinite or semi-infinite cable theory was found to be generally applicable. For above-ground cables, the peak current saturates when cable lengths are longer than the drive pulse length times the speed of light ( $\sim$ 76 m in this case). The semi-infinite cable formulas of reference 1 thus apply for the 100 and 1000-m cables. Note also that the finite ground conductivity ( $\sigma$ ) makes the response of the cable near the ground larger than it would be near a perfect conductor.

For the 10-m cable above the ground, the formulas in reference 1 are not applicable. Estimates for this case were thus based on simple capacitive coupling  $(1 \sim C\dot{V})$ . Note that this type of current estimate increases linearly with cable length and is thus invalid for the longer cables.

For buried cables, infinite cable theory is applicable for lengths greater than  $\ell$ , where

 $\ell > 10^9 \sqrt{\tau \tau_e}$ ,  $\tau = \text{pulse width} \sim 250 \text{ ns},$  $t_e = \epsilon_0 / \sigma \simeq 10^{-9} \text{ s}.$ 

In this case  $\ell \simeq 15$  m. Such theory was applied for all three cable lengths.

Note that characteristic current pulse widths correspond to the shorter of the drive pulse widths or to the cable length divided by the speed of propagation down the cable,  $\sim 0.25$  ms.

The effects of soil attenuation have been taken into account by using 1 MHz as the mid-frequency of the energy contained in the double exponential pulse. The attenuation  $(A_S)$  is then

where  $A_{S} = \exp[-d\sqrt{\pi f \mu_{0} \sigma}] \simeq -d \frac{f_{0}}{503};$  $\mu_{0} = 4\pi \times 10^{-7} \text{ H/m},$ d = depth in metres,f = frequency (Hz), $\sigma = \text{ground conductivity (mho)}.$ 

The magnitudes of the vulnerability levels, as presented in table 8, were then calculated from

$$|E| (V/m) = \frac{\log^{-1} \frac{SE}{10}}{K A_S} \sqrt{\frac{\overline{W}_{threat}}{Z_{tpw}}} .$$

vhere	s <sub>E</sub>	= shielding effectiveness of cables, 80 dB and 40 dB assumed,
	К	= peak current coupled onto cable (ref 8, p 86),
	AS	= attenuation of the earth at the burial depth,
	W <sub>threat</sub>	= energy level of the threat $(5 \times 10^{-6} \text{ J for burnout and} 1 \times 10^{-9} \text{ J for upset})$ ,
	Z	= termination impedance of the cable,
	tpw	= effective pulse width (250 ns for cables longer than 76 m and 40 ns for shorter cables).

#### **RADIATION EFFECTS ON ELECTRONIC SYSTEMS**

The radiation environments encountered by military systems are classified as nuclearweapon-generated and natural (or space). These classifications may be further divided into components that produce basic effects on the electronics in the system. Many variations and combinations of the primary components of these two classifications exist and depend upon the particular system applications. The problem for the system designer is to achieve costeffective and balanced hardening against all components of the particular radiation environment to ensure the survivability of the system. To accomplish this goal, he must be familiar with the primary components of the radiation environments and their effects on the system electronics. A brief description of the primary components of the two radiation environments, along with their basic effects, is presented in table 9 and the following discussion. The sources of the primary components in the radiation environments have been described in several documents, and are beyond the score of this document.

The primary concern of system designers when hardening the system to radiation effects is for the semiconductors (integrated circuit and discrete) incorporated in most modern military systems, since they are the most vulnerable electronic components. This is not to say that there are no radiation effects in the passive components (eg, resistors, inductors, capacitors), but that they are typically less susceptible. However, in those military systems that must survive very severe radiation environments these components, especially electrolytic capacitors, must be included in the overall hardening effort.

The response of the system to the radiation environment may be categorized according to whether the effect is transient or permanent. In this case, "transient" and "permanent" refer to the effect produced by the radiation, not to the duration of the radiation environment. Transient effects in an electronic system may be produced by two primary components of the radiation environment: energetic particles penetrating the semiconductor materials, and EMP-generated signals coupled into the system. The first effect creates electron-hole pairs which produce photocurrents in the devices. This process results in ionization effects. The magnitude and duration of the effect depend on the number of ionizing particles that interact with the semiconductor material, its active volume, the number of electron-hole pairs created by each particle, and how long it takes them to either recombine in or exit the device. The observed response of an electronic circuit could range from an increase in circuit noise to large currents  $(10^{-3} \text{ to } > 10 \text{ A})$ flowing in the circuits.

Permanent changes within semiconductors may also be produced by many of the components of the radiation environment. These are usually categorized as bulk, surface, thermomechanical, and EMP-related effects. Bulk effects are produced by those constituents of the radiation environment (neutrons, protons, electrons, etc) that produce permanent

Radiation	Units	Basic Effect	
Natural			
Electron fluence	e/cm²	lonization of surface material. Causes degradation deterioration, charge buildup on satellite surfa components.	
Proton fluence	p/cm²	Permanent degradation in solar cells and other directly exposed semiconductor devices.	
Weapon			
X-ray			
Thermomechanical	cal/cm <sup>2</sup>	Mechanical deterioration in form of spallation, glazing, cracking, weakening of mechanical integrity.	
Ionization (prompt)	rd(Si)/s	Induced photocurrents cause transient upset and high currents in all electronics. Potential latchup of junction-isolated ICs.	
Gamma-ray ionization (prompt)	rd (Si)/s	Same as X-ray prompt ionization effects.	
Total ionizing dose	rd (Si)	Total accumulated ionizing radiation causes permanent changes in semiconductors. MOS is most susceptible.	
Neutron fluence	n/cm²	· Permanent displacement in semiconductor mate- rial lattice structure, causing part degradation.	
EMP and system- generated EMP (SGEMP)	V/m	SGEMP produced by X-ray environment. Both EMP and SGEMP induce currents in intercon- necting cables, in antennas, and throughout a system.	
Blast and shock		Creates overpressure that causes mechanical deformation and creates winds that pick up debris that can cause further damage.	

Taken from reference 8

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Table 9. Primary components of radiation environments

<sup>8</sup> IRT Report 4521-002, The ABC's of Radiation Hardening, April 1976

displacements within the crystalline structure of the semiconductor materials. The displacements will alter many of the parametric characteristics of the semiconductor devices such as forward current transfer ratio ( $h_{FE}$ ), leakage currents, and capacitances.

The ionization components of the radiation environment may create permanent and transient parametric changes in the semiconductors. The principal permanent ionization effects are surface effects and burnout.

Surface effects are primarily due to trapped charges at or near the surfaces of the semiconductor active areas. Some parametric changes within semiconductor devices that may be caused by surface effects are in transconductance, capacitances, and leakage currents. If the ionization components of the radiation environment have sufficient magnitude to cause large current densities within the semiconductors, enough energy can be dissipated in the active areas of the devices to cause permanent parametric changes or, as a worst case, catastrophic failure (burnout).

The X-ray components of the radiation environment also may produce permanent effects on the system electronics. When low-energy X-rays are absorbed by a material, they deposit energy during the pulse in the form of heat. This creates a shock wave within the material, and if sufficient energy is released the stress buildup in the material will result in physical damage. The absorption and propagation of the energy is highly dependent on the material, but is usually larger in those materials with high atomic number (high Z). The heat produced may also be sufficient to cause spallation of the material.

A summary of the transient radiation effects in electronics is presented in figure 1, which shows a spectrum of typical nuclear effects on electronic systems for the neutron fluence, dose-rate, and total-dose environments. The impact of these effects on a system will depend heavily on the susceptibility of the circuit and its function. For example, in a missile application, the occurrence of an upset of a system may be critical to mission success, while in a satellite, a system upset could be nothing more than noise in the system.

Considerable attention has been given to system responses associated with the EMP threat from a nuclear weapons environment, since the incident fields generated by a single optimally positioned nuclear detonation may cover very large areas of the earth's surface. Systems that are most vulnerable to this threat are those that present large coupling volumes for the incident electromagnetic energy, such as hard-wired communication networks and aircraft. These systems, if not sufficiently protected, can couple enough electrical energy into the semiconductor piece parts to cause burnouts or upsets in those devices that directly interface with the hard-wired interconnections.

The coupling paths for fields external to the system are numerous. They include diffusion through the enclosure outer walls or leakage through penetrations, either deliberate or inadvertent. Deliberate penetrations include antennas, radars, and cable feedthroughs; inadvertent penetrations include doors, windows, bomb bays, and other apertures that are less obvious such as hydraulic lines and power-line pickups. In aircraft, this could occur through generators near air intakes, and, for permanent ground-based installations, the commercial power distribution networks.

The amount of electromagnetic leakage through various penetrations depends on the type of penetration and the total fields near the penetrations. Some penetrations are more sensitive to the local electric field, others to magnetic fields. Also, at some locations on the exterior, total electric fields will be large and magnetic fields small, while the opposite may be true elsewhere. This will depend on the geometry of the enclosure and its orien-

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Neutron Effects	Dose-rate Effects	' Total-dose Effects
(n/cm <sup>2</sup> )	[rd(Si)/s]	[rd(Si)]
10 <sup>15</sup> – MOSFET degrades Hardened logic degrades	10 <sup>11</sup> - Semiconductors saturated completely; currents unlimited	10 <sup>6</sup> - ECL degrades Some degradation in most semiconductors
10 <sup>14</sup> – TTL, RF, & FET transistors degrade	10 <sup>10</sup>	10 <sup>5</sup> – I <sup>2</sup> L degraded, hardened CMOS & other MOS show degradation
Most transistors & RTL/DTL degrade		Power transistors show degradation
10 <sup>13</sup> Low-frequency transistors degrade	10 <sup>9</sup> – Hardened logic threshold Transistors turned on hard	10 <sup>4</sup> – Commercial PMOS/ CMOS gate shifts
10 <sup>12</sup>	10 <sup>8</sup> – Potential latchup condition	10 <sup>3</sup> – First commercial CMOS gate shifts
Power transistors 10 <sup>11</sup> degrade SCR & UJT	Significant I <sub>pp</sub> * in most semiconductors 10 <sup>7</sup> –	10 <sup>2</sup> — Shift in quartz crystal frequency (∿2 ppm)
degrade 10 <sup>10</sup>	Photocurrent in PIN 10 <sup>6</sup>	10 <sup>1</sup> -

\*Ipp = Photocurrent magnitudes

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Figure 1. Spectrum of radiation effects on semiconductor components and estimated susceptibility ranges (ref 1)

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tation to the incident EMP and radiation.

For systems that are close enough to the detonation to receive significant photon energies, other EMP threats are generated within the system. These are system-generated EMP (SGEMP) and internal EMP (IEMP).

System-generated EMP is a result of the low-energy photon flux interacting with the external structural enclosures of the system. The low-energy photon interactions produce secondary-electron emission currents that create external surface currents and charge densities which contribute to the total external electric and magnetic fields. These total fields may then be coupled into the internal cables and circuits through the previously mentioned penetrations or apertures.

High-energy photons will penetrate  $1^{1/2}$  external walls of the system and create secondary-electron emissions from the internation faces of the enclosures and shielding components. The generated internal electrical and magnetic fields are then the source of the internal EMP threat. The magnitude of the generated fields is dependent on the incident photon flux and its spectrum, the size and shape of the enclosures, and the materials from which they are constructed.

In a ground based system near a nuclear attack, the source-region nuclear effects present formidable problems to the designer. A significant problem is the protection of equipment located at a sufficient distance to survive the direct attack but connected to long hard-wired cables in the vicinity of the detonation. Peak currents on the shields of these cables on the order of 100 kA that last for  $\sim 30 \ \mu s$  have been estimated by several theoreticians. For a typical 50-ohm coaxial cable with a shielding effectiveness of 80 dB, the energy on the center conductor at its terminations would be 15 mJ. Providing reasonable protection against this threat is difficult.

#### NUCLEAR VULNERABILITIES OF ELECTRO-OPTIC COMPONENTS

Electro-optic techniques can provide definite advantages to military systems over conventional techniques for gathering, storing, processing, and transmitting signals and data. Among electro-optic techniques is fiber optics technology. Proven benefits to the military of systems employing this technology include increased bandwidths, data rates, and transmission distances without repeaters, and various combinations of reduced weight and power consumption. Possible benefits, indicated by studies but not yet adequately demonstrated, include higher reliability, lower maintenance and life-cycle costs, longer mean time between failures, and greater EMP hardening.

It is the latter of these that has been the driving force for many radiation-hardening studies. The rationale for these studies has been that, since the fiber optic cable is immune to em radiation due to its dielectric nature, the system's hardness to EMP should be increased. This is logical for systems that consist of only fiber optic cables and are exposed to only the em energy components of the radiation environment. However, the real world requires that fiber optic systems have electronics (which are sensitive to the nuclear environment) associated with the cable, and relatively few military systems have only the em radiation as the total nuclear threat. Therefore, this section of the report presents a discussion of the response of fiber optic systems to the total radiation environment.

At the present state of development in fiber optic technology, lack of component standardization precludes easy interchange of the electro-optical components comprising the system. Therefore, when selecting these components, a system-level point of view must

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be used; components may not be selected from just a radiation-hardness point of view because of the dependence of one component on another in overall system performance. The selection process is usually accomplished by a very long, complex, and iterative procedure that involves tradeoffs between optimum electro-optical performance parameters and optimum radiation hardness parameters. To aid in this procedure, the designer usually uses a link power or link loss budget, which may be a table or formula that presents the optical powers (or losses) at the electro-optical component interface points, along with the system requirements. The resultant calculated optical power, either excess or deficiency, is referred to as the design margin and is a measure of how well the system meets performance specifications.

The design margin may also be used as a measure of the system's radiation hardness to those effects that cause permanent or slow-recovery transient degradation in the component's operating parameters. However, use of only the design margin to predict transientupset radiation hardness may lead to erroneous conclusions. Hardness to transient upset is dependent on both the design margin and the magnitude of received optical signals. The reason for this may be easily understood if one considers how the design margin may be increased by increasing either receiver sensitivity or received optical power. Measures taken to increase receiver sensitivity will usually result in an increase in its radiation sensitivity which, in turn, results in a lower radiation-upset threshold. Therefore, from a radiation hardness point of view it is more advantageous to increase the system's received optical power, if possible. However, tradeoffs exist and other system design parameters, such as reliability and operating lifetime, may be adversely affected by methods taken to obtain the increase in optical power.

To enable the reader to understand the complexity of problems faced by the designer in selecting electro-optic components for use in a fiber optic system, a brief discussion of the pertinent radiation response of the components will be presented and trade-offs affecting the overall design of the fiber optic system will be identified. For purposes of this discussion, the fiber optic system will be divided into seven sections – drivers, sources, connectors, fibers, detectors, amplifiers, and (for some systems) fiber optic couplers. Information presented here was taken from a report prepared for DNA under contract DNA001-76-C-0139 by IRT Corporation. The material contained in that report has been updated where new data were available.

#### DRIVERS

Electronic integrated and hybrid circuits designed specifically for driving solid-state light sources are being developed by some IC manufacturers. These circuits are intended primarily for digital applications. Some have been developed for the military under government contract, but no radiation-hardening requirements had been placed on them. Therefore, it is expected that these circuits will be no more radiation-resistant than the standard IC processing used in their construction. Fortunately, for most applications the speed of operation required to meet the electrical operating specifications for these circuits is high enough to require that high-frequency transistors be used in their construction. To obtain high-frequency response, manufacturers must use small-geometry, high  $f_t$  semiconductors in constructing the low-level signal-handling portions of these ICs. These areas are thus made less sensitive to the ionization components and the displacement damage-producing components of the radiation environments. However, the semiconductors that perform the buffer/driving function to the light source must handle much larger currents (10 – 200 mA) and, therefore, must have larger junction areas to reduce the power dissipation in the devices

and to increase thermal conductivity to the packages. The larger geometry presents larger radiation-sensitive volumes which will produce greater photocurrent responses and will probably set the upset levels for these devices.

Only one type of IC driver (SPX 3619, built by Spectronics, Inc) has been radiationtested and reported in the literature (ref 9). The responses in a high-dose-rate, ionizing radiation environment indicated that the output transistors contributed to most of the photocurrent responses of the IC and that no significant (10%) permanent degradation occurred while in any of the radiation environments in which they were tested [~1 Mrd(Si) total dose from combined gamma, electron, and proton environments and  $3 \times 10^{12} n/cm^2$ neutrons].

Primary radiation environmental concerns for the systems designer, over which he has no control, are whether the output drive current switching level and frequency response will degrade with accumulation of ionization total dose or with displacement damage-producing components of the radiation environments. Assuming that the light source is not an integral part of the driver circuitry, the designer may protect it from excessive drive currents produced in the driver IC when subjected to high-dose-rate ionization environments. This may be accomplished by using external current-limiting techniques such as resistors or active bypass networks. A possible EMP vulnerability exists with these devices because they require electrical signals wired to their inputs and receive their power via the power supply bus. If these hard-wired connections couple enough energy into the device, burnouts or upsets will be induced within the devices. The sensitivity of these devices will be no greater than that of other devices developed by the same technologies. The range of thresholds will then be ~10 to 100  $\mu$ J for burnouts and ~1 to 10 nJ for upset levels (ref 10-12).

## LIGHT SOURCES

This report treats only the two types of semiconductor light sources most likely to be used by the military in fiber optic communications systems. These are light-emitting diodes (LED) and laser diodes, both of which can be fabricated from many different III-V semiconductor materials to give somewhat different properties, primarily different emission wavelengths. The wavelength chosen must be compatible with the transmission properties of the fiber and ranges from 0.6 to  $\sim 1.3 \,\mu m$ . The tendency at present is toward development of sources that emit in the 1.0 to  $1.3 - \mu m$  range since some of the more promising fibers have lower optical attenuation and possibly less radiation response at these wavelengths.

<sup>&</sup>lt;sup>9</sup> IRT Final Report for AFWL, Fiber Optic Radiation Test and Evaluation Program, Kirtland AFB, December 1978

<sup>&</sup>lt;sup>10</sup> Motorola Application Note AN-707, Noise Immunity Comparison of CMOS vs Popular Bipolar Logic Families, by AA Allen, 1973

<sup>&</sup>lt;sup>11</sup> Wunsch, DC, and Bell, RR, Determination of Threshold Failure Levels of Semiconductor Diodes and Transistors Due to Pulse Voltage, IEEE Trans Nucl Sci NS-17, 364 1970

<sup>&</sup>lt;sup>12</sup> Tasca, DN, Pulse Power Failure Modes in Semiconductors, IEEE Trans Nucl Sci NS-17, 364, 1970

Alloy combinations such as InGaAs, InGaP, InGaAsP, and GaAlSnSb have been used to construct these emitters, but the most serious efforts are presently centered around InGaAsP. Another reason for this developmental trend is that the pulse dispersion of optical signals in the fibers is less at these wavelengths. This is significant because it makes possible longer transmission distances without repeaters at higher frequencies of operation.

The effects that radiation can have on optical sources and the general range of thresholds for observation of these effects are presented in table 10.

Degradation Level	Specific Effect	Optical Source		
		LED	Laser Diode	
Catastrophic Damage	Ionization-induced burnout	10 <sup>10</sup> rd(Si)/s	10 <sup>9</sup> rd(Si)/s	
	Electrical pulse burnout (100 ms)	50 - 3000 μJ	0.5 <b>-</b> 30 µJ	
Transient Ionization Effects	Transient upset	$3 \times 19^9$ rd(Si)/s	10 <sup>12</sup> - 10 <sup>13</sup> rd(Si)/s	
Permanent Degradation	Light output loss, ionization	10 <sup>7</sup> rd(Si)	10 <sup>7</sup> rd(Si)	
	Light output loss, neutron	$10^{12} - 10^{14} \text{ n/cm}^2$	$10^{13} - 10^{15} \text{ n/cm}^2$	
Thermomechanical Effects	Mechanical failure	l cal/g(Au)	l cal/g(Au)	

Table 10. Damage thresholds of optical sources.

## **Catastrophic Damage**

The ionization-induced threshold of  $10^{10}$  rd(Si)/s for LEDs is a conservative figure since this was the maximum test level in the only experiments reported (ref 9,13) and only a few devices failed. The threshold may be lower for laser devices because they are operated at higher current densities and have smaller junction areas. The  $10^9$  rd(Si)/s listed in table 10 is just a guess, but it is conservative since it is an order of magnitude lower than the LED threshold.

The electrical pulse burnout thresholds where the devices failed were measured directly in certain GaAs, GaP, and GaAs<sub>x</sub>P<sub>1-x</sub> devices (ref 14). These devices generally are fast and, thus, are small-area devices. According to the Wunsch-Bell/Tasca model (ref 11,12) for burnout, the threshold is area-dependent, as was found to be true over the limited area range of the tested devices. Hence, the thresholds for the larger-area, high-power LEDs and the smaller-area laser diodes were scaled from the measured data by the area ratio. The GaP device tested has a large area and was a slow unit. Faster units of this type may have lower thresholds. The Ga<sub>x</sub>A1<sub>1-x</sub>As LEDs were assumed to be similar to GaAs<sub>x</sub>P<sub>1-x</sub> LED.

Laser diodes are small-area devices, and since the threshold is proportional to area, they are predicted to be more vulnerable than LEDs. Experimental information is almost

<sup>&</sup>lt;sup>13</sup> AFWL TR-74-302, An Assessment of Fiber Optics Technology for Satellite Hardening, by AH Kalma, 1975

entirely lacking in this field, except for an operating-life test which showed that  $Ga_xAl_{1-x}As$  cw devices fail at about 1  $\mu J$  for 100-ns pulses (ref 15). This agrees reasonably with the estimate in the table and reinforces its credibility. Since the estimated thresholds are low, this area is a very important one in which experimental information is needed.

#### **Transient Ionization Effects**

Transient upset causes a false signal that decays rapidly after the burst. There have been almost no studies of ionization-induced emission in LEDs or laser diodes. This is because the devices are operated at forward bias with high current densities, and photocurrents would be relatively less important. For estimation purposes, the photocurrent response of GaAs diodes is used. This value is  $\sim 3 \times 10^{-9} (A/cm^2)/[rd(Si)/s]$  (ref 16). The predictions agree reasonably with results from the reported experiments on devices made from GaAs, GaP, GaAs<sub>x</sub>P<sub>1-x'</sub>and Ga<sub>x</sub>A1<sub>1-x</sub>As (ref 3, 13).

Another transient ionization effect is the light produced by the steady-state ionization encountered in the space environment or by the delayed gammas from a prompt burst. In light sources this would produce a background light level which would be similar to added noise. This effect would remain while the radiation remained and disappear when the radiation was not present. There are no experimental data in this area; the numbers shown in the table are predictions from the photocurrent response relationship. Again, laser degradation thresholds are much higher than LED thresholds because of their higher operating current density.

#### **Permanent Degradation**

Permanent ionization effects are produced by the total ionizing dose absorbed by a device. Surfaces or interfaces are affected primarily through increased leakage current, with some change in capacitance. The light sources are essentially unaffected by leakage current, and no ionization mechanisms (as opposed to displacement damage mechanism) have been reported (ref 3, 17, 18). Even though it is displacement damage that causes the degradation, most studies were performed with gamma fluence (ref 19-28), so this is how the thresholds are shown in table 10.

Displacement effects in the light sources produce competing nonradiative recombination centers and degrade the light output of the LEDs. In laser diodes a more important process is the lasing threshold shift. Lasers operated well above threshold will not show much degradation at low fluences. Eventually, the threshold shifts enough that the output drops drastically. The fluence required for this to occur in devices operated at a power factor of 3 from the maximum is considered the threshold.

The thresholds shown in table 10 are taken from a large number of studies (ref 23-42). The range of thresholds is apparently due to varying device quality, with the better-quality devices being more vulnerable. The better-quality devices have higher efficiency and, thus, fewer initial nonradiative recombination centers; therefore, it requires a lower radiation fluence to introduce a comparable number of new centers.

14-42 (see p 28, 29)

- <sup>15</sup> O'Brien, JT, et al, Recent Developments in Injection Lasers, presented at Conference on Laser and Electro-Optical Systems, San Diego, California, 25-27 May, 1976
- <sup>16</sup> AFWL TR-68-31, vol 2, Radiation Effects on GaAs Devices and Schottky Diodes, by HD Southward and RH Schnurr, August 1968
- <sup>17</sup> Northrop Report DNA3776F, Study of Radiation Effects in Bulk CMOS Microcircuits, IIL/LSI Logic Cells, and Optic Couplers, by JP Raymond, et al, 1975
- <sup>18</sup> Barnes, CE. Development of Efficient Radiation-Insensitive GaA:Zn LEDs, IEEE Trans Nucl Sci NS-24, No. 6, December 1977
- <sup>19</sup> AFAL TR-74-61, Survey of Vulnerability of Contemporary Semiconductor Components to Nuclear Radiation, by RP Donovan, et al, 1974
- <sup>20</sup> Epstein, AS, et al, Gamma Irradiation and Annealing Effects in Nitrogen-Doped GaAsP Green and Yellow Light-Emitting Diodes, IEEE Trans Nucl Sci NS-19, 386, 1972
- <sup>21</sup> Barnes, CE, Effects of <sup>60</sup>Co Gamma Irradiation on Epitaxial GaAs Laser Diodes, Phys Rev B1, 4735, 1970
- <sup>22</sup> Schnurr, RH, and Southward, HD, Radiation Effects on GaAs Devices, IEEE Trans Nucl Sci NS-15, 306, 1968
- <sup>23</sup> Soda, KJ, et al, The Effect of Gamma Irradiation on Optical Isolators, IEEE Trans Nucl Sci NS-22, 2475, 1975
- <sup>24</sup> Polimadei, RA, et al, Performance of GaA1As Light-Emitting Diodes in Radiation Environments, IEEE Trans Nucl Sci NS-21, 96, 1974
- <sup>25</sup> Share, S, et al, Radiation Effects and Hardening on Compensated GaAs Light-Emitting Diodes, IEEE Trans Nucl Sci NS-20, 256, 1973
- <sup>26</sup> Epstein, AS, et al, Radiation Effects on Active and Passive Optical Components, presented at DoD/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- <sup>27</sup> Schroeder, JD, et al, Radiation-Damage-Induced Time Delay in GaAs Lasers, IEEE Trans Nucl Sci NS-19, 391, 1972
- <sup>28</sup> Epstein, AS, and Trimmer, PA, Radiation Damage and Annealing Effects on Photon-Coupled Isolators, IEEE Trans Nucl Sci NS-19, 391, 1972
- <sup>29</sup> Barnes, CE, Radiation Effects in Electroluminescent Diodes, IEEE Trans Nucl Sci NS-18, 322, 1971
- <sup>30</sup> Noel, BW, et al, Neutron Irradiation Effects on Diffused GaAs Laser Diodes, IEEE Trans Nucl Sci NS-18, 378, 1971
- <sup>31</sup> Barnes, CE, Neutron Damage in Epitaxial GaAs Laser Diodes, J Appl Phys 42, 1941, 1971
- <sup>32</sup> Schade, H, et al, Defect Centers in GaAsP Electroluminescent Diodes Due to High-Energy Electron Irradiation, J Appl Phys 41, 3783, 1970

<sup>&</sup>lt;sup>14</sup> Kalma, AH, and Fisher, W, Electrical Pulse Burnout Testing of Light-Emitting Diodes, IEEE Trans Nucl Sci NS-22, 2510, 1975

The  $Ga_xAl_{1-x}As$  devices are relatively new and have not been heavily tested. Where they have been (ref 3), they appear to be comparable to  $GaAs_xP_{1-x}$  devices, and both these types seem to be an order of magnitude harder than GaAs devices. In fact, use of ternary compounds to fabricate devices has been proposed as a hardening technique (ref 18,25,26). At times, the limited testing of the ternary material devices has produced degradation thresholds only at one end of the expected range. Some early GaAs work indicated neutron degradation thresholds higher than those listed in the table. The tested devices probably were of lower quality than those currently available, and this resulted in their being less vulnerable.

#### **Thermomechanical Effects**

No data exist concerning thermomechanical damage in light sources. The thresholds listed in table 10 assume that the light sources behave similarly to silicon semiconductor devices. This means that the contact bonds are the most vulnerable point of the devices. The light sources will almost assuredly be protected from the direct X-ray beam, so it is unlikely that any of the damaging low-energy X-rays would reach them.

#### CABLES

Table 11 presents the damage thresholds for fibers. Almost all effects depend on fiber length, so the thresholds are quoted per unit length. To calculate the thresholds, it has been assumed that in most receivers the noise level would be equivalent to an input signal of a few nanowatts. Thus, the minimum signal required for a 20 dB S/N ratio would be about 100 nW, and this is assumed to be the conservative signal leaving the fiber.

<sup>&</sup>lt;sup>33</sup> Barnes, CE, Neutron Damage in GaAs Laser Diodes at and above Laser Threshold, IEEE Trans Nucl Sci NS-19, 382, 1972

<sup>&</sup>lt;sup>34</sup> Epstein, AS, et al, Effect of Neutron Irradiation on GaAsP Electroluminescent Diodes, Appl Phys Lett 23, 472, 1973

<sup>&</sup>lt;sup>35</sup> Stanley, AG, Comparison of Light-Emitting Diodes in a Space Radiation Environment, IEEE Trans Nucl Sci NS-17, 239, 1970

<sup>&</sup>lt;sup>36</sup> Sandia Report SC-RR-710686 Radiation Effects in Light-Emitting Diodes and Laser Diodes, by CE Barnes, 1972

<sup>&</sup>lt;sup>37</sup> Barnes, CE, Neutron Damage in GaP Light-Emitting Diodes, Appl Phys Lett 20, 110, 1972

<sup>&</sup>lt;sup>38</sup> Compton, DMJ, and Cesena, RA, Mechanism of Radiation Effects on Lasers, IEEE Trans Nucl Sci NS-14, 55, 1967

<sup>&</sup>lt;sup>39</sup> Arnold, GW, Luminescence in Intrinsic and Annealed Electron-Irradiated GaAs:Cd, Phys Rev 183, 777, 1969

<sup>&</sup>lt;sup>40</sup> Barnes, CE, Appl Phys Lett 20, 86, 1972

<sup>&</sup>lt;sup>41</sup> Barnes, CE, Tunneling in SiC Electroluminescent Diodes, J Appl Phys 45, 193, 1974

<sup>&</sup>lt;sup>42</sup> Barnes, CE, Increased Radiation Hardness of GaAs Laser Diodes at High Current Densities, J Appl Phys 45, 3485, 1974

Degradation Level	Specific Effect	Special Glass-Clad Silica Fibers	Polymer-Clad Silica Fibers	Plastic Fibers
Transient Ionization Effects	Transient upset signal	10 <sup>6</sup> - 10 <sup>8</sup> rd(Si)/s-m	10 <sup>4</sup> - 10 <sup>6</sup> rd(Si)/s-m	10 <sup>4</sup> - 10 <sup>5</sup> rd(Si)/s-m
	Transient signal loss (immediate)	10 <sup>10</sup> rd(Si)/s-m	3 × 10 <sup>10</sup> rd(Si)/s-m	3 × 10 <sup>10</sup> rd(Si)/s-m
	Transient signal loss (1 ms)	10 <sup>9</sup> - 10 <sup>11</sup> rd(Si)/s-m	3 × 10 <sup>11</sup> rd(Si)/s-m	10 <sup>11</sup> - 10 <sup>12</sup> rd(Si)/s-m
Permanent Degradation	Ionization- induced signal loss	$10^{0} - 10^{3}$ rd(Si)/s-m	10 <sup>0</sup> - 10 <sup>3</sup> rd(Si)/s-m	10 <sup>5</sup> - 10 <sup>7</sup> rd(Si)/s-m
Thermomechanical Effects	Mechanical failure	10 cal/g(Au)	10 cal/g(Au)	10 cal/g(Au)

Table 11. Damage thresholds of fibers

## **Catastrophic Damage**

Optical fibers, having no junctions, are immune to this effect. They are also immune to electromagnetic pickup, and therefore, do not produce electrical pulses capable of burning out other components. This is the primary reason that fiber optic systems are considered for radiation hardening.

## **Transient Ionization Effects**

Transient ionization effects are the most important radiation vulnerability mechanisms in fibers, and most of the studies of radiation effects in fibers have examined this point (ref 43-60).

- <sup>43</sup> Evans, BD, and Sigel, GH, Jr, Radiation-Resistant Fiber Optic Materials and Waveguides, IEEE Trans Nucl Sci NS-22, 2462, 1975
- <sup>44</sup> Mattern, PL, et al, Effects of Radiation on Absorption and Luminescence of Fiber Optic Waveguides and Materials, IEEE Trans Nucl Sci NS-21, 81, 1974
- <sup>45</sup> Evans, BD, and Sigel, GH, Jr, Permanent and Transient Radiation-Induced Losses in Optical Fibers, IEEE Trans Nucl Sci NS-21, 113, 1974
- <sup>46</sup> NRL Memorandum Report 2704, Radiation Effects in Fiber Optic Waveguides, by GH Sigel, Jr, 1973
- <sup>47</sup> Mattern, PL. Radiation-Induced Absorption and Luminescence in Glass and Plastic Optical Waveguides, presented at DoD/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- <sup>48</sup> NRL Memorandum Report 2934, Radiation Effects in Fiber Optic Waveguides, by GH Sigel, Jr, et al, 1974

Two transient effects in fibers can be produced by ionization. One is luminescence, which would produce an upset signal. The second is an increase in attenuation. The latter is the most important radiation effect in fibers because it decays slowly with time following exposure. Slow decay means that the transmission capability of a link will be lost not only during exposure but possibly for a long period following exposure. The thresholds shown in table 11 assume a 100-ns pulse width. This arbitrary choice is in the area of pulse widths at which exposure may occur. Scaling to other pulse widths should be done by the ratio of pulse widths.

The initial high absorption can produce a transient loss of signal if the dose is high enough. The threshold depends on the time interval after the burst that is required for the system to return to operation. The thresholds shown are for continuous operation ("immediate"), for operation 1 ms after the burst, and for operation 1s after the burst. These times are arbitrarily chosen to show how the threshold is affected. Recovery of the plastic fibers takes longer in a vacuum than in room-ambient conditions, apparently because of some atmospheric constituent (probably oxygen) scavenging the absorption centers (ref 50,52-54). The range of thresholds shown for transient upset signal depends on whether the data were measured using X-rays or low-energy electrons, with electrons providing the lower threshold.

- <sup>50</sup> AFCRL-TR-75-0190, Radiation Effects on Fiber Optics, by JA Wall and JF Bryant, 1975
- <sup>51</sup> Mattern, PL, et al, Effects of Radiation on Optical Fibers and Materials, IEEE Trans Nucl Sci NS-22, 2468, 1975
- <sup>52</sup> Sandia Report SAND76-8056, A Summary of Radiation-Induced Transient Absorption and Recovery in Fiber Optic Waveguides, by CD Skoog, 1976
- <sup>53</sup> IRT Report INTEL-RT 8111-089, Fiber Optics Test Program for Satellite Applications, by AH Kalma and WH Hardwick
- <sup>54</sup> Kalma, AW and Hardwick, WH, Radiation Testing of a Fiber Optics Data Link, IEEE Trans Nucl Sci NS-23, 1769, 1976
- <sup>55</sup> Maurer, RD, et al, Effect of Neutron and Gamma Radiation on Glass Optical Waveguides, Appl Opt 12, 2024, 1973
- <sup>56</sup> Sigel, GH, Jr, and Evans, BD, Effects of Ionizing Radiation on Transmission of Optical Fibers, Appl. Phys Lett 24, 410, 1974
- <sup>57</sup> Acharekar, MA, Multimode Radiation-Resistant Glass Fiber for Data Transmission Applications, presented at DoD/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- <sup>58</sup> Sigel, GH, Jr, Radiation Damage in Fiber Optics, presented at DoD/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- <sup>59</sup> Friebele, EJ, et al, Radiation Protection of Fiber Optics Materials: Effects of Oxidation and Reduction, Appl Phys Lett 24, 412, 1974
- <sup>60</sup> Friebele, EJ, et al, Defect Centers in a Germanium-Doped Silica-Core Optical Fiber, J Appl Phys 45, 3424, 1974

<sup>&</sup>lt;sup>49</sup> AFCRL-TR-74-0012, Transient Radiation Effects Tests of a Corning Radiation-Resistant Optical Fiber, by JA Wall, 1975

Steady-state transient effects in fibers are much the same as the above pulse effects. These could be produced by either delayed gammas or space electrons, both of which produce ionization for much longer periods of times than the prompt pulse, hence, the term "steady state". The most important effect is signal loss which does not occur at the same dose rate as for prompt transient pulses because recovery is occurring at the same time as generation. To predict the amount of attenuation produced by a steady state environment it is necessary to combine the production rate and the recovery rate, with the time elapsed since any exposure increment (ref 3-6, 50, 53, 54). The threshold depends on the exposure time and the magnitude of the steady state flux. The thresholds shown in the table assume a 10-s exposure to a constant flux, and require that the attenuation be low enough for link operation at the end of this period. Some data exist on the steady state response of fibers (ref 3-6,53,54), but none on the performance of a fiber optic system; therefore, the thresholds are the result of calculations.

A second effect of a steady state environment is the increased noise due to increased background light, which would be the result of luminescence. The increased noise could degrade the available S/N level of a fiber optic link, but it has not been reported.

#### **Permanent Degration**

The ionization-induced permanent degradation is the increase in fiber absorption which decreases the transmitted signal (ref 13,43-60). The absorption is the same effect as the transient absorption described above. Although the increased absorption decays indefinitely, the portion considered permanent degradation is that present one or two days after exposure.

Displacement damage cannot be measured in the fibers because the accompanying ionization produces more damage than displacement does (ref 13,43,49-52,55). The fibers are, for the purposes of this study, immune to displacement damage.

#### **Thermomechanical Effects**

The most likely effect of thermomechanical damage in fibers is surface cracking produced by the heating (ref 61) and this is the threshold shown. It is a higher threshold than for similar effects in devices because there are no wire bonds in the fibers; these bonds are the weak points. Another effect in the fibers could be dielectric breakdown caused by fiber charging. Calculations indicate that the threshold for breakdown is of the same order of magnitude as that for cracking caused by heating in the extreme case of using a gold converter, assuming that all the electrons are absorbed at the surface of a plane of fiber material with no leakage. Since this is worse than would occur in practice, dielectric breakdown would probably not be a vulnerability problem for an X-ray threat. However, for the natural or weapon-enhanced space environments this could be a problem.

<sup>&</sup>lt;sup>61</sup> Baba, A, et al, Effect of Simulated Nuclear Thermal Pulses on Fiber Optic Cables, IEEE Trans Nucl Sci NS-26, No 6, December 1979

## DETECTORS

Limited studies (ref 4,62) of the radiation effects in photodetectors have been made, but not with any degree of completeness, and unfortunately many of the threshold levels must be inferred. The information is summarized in table 12.

In determining upset and noise thresholds, some knowledge of the input light level and system operation, including system noise and bandwidth, is required. On the pessimistic side, it is assumed that the noise level is not set by the photodetector itself but by the amplifier, where the noise is about an order of magnitude higher in well designed systems. System bandwidth would also be somewhat less. Therefore, the listed thresholds of diode devices are low by at least this order of magnitude in systems using currently available electronics. However, improvements could lower the noise of the electronics and make the detector noise more important. Thus, the listed thresholds are minimum ones for a system that may be approached in the future.

Degradation Level	Specific Effect	Pin Diode	Avalanche Photodiode
Catastrophic Damage	Ionization-induced burnout	10 <sup>9</sup> rd(Si)/s	10 <sup>9</sup> rd(Si)/s
	Electrical pulse burnout (100 ns)	100-40,000 μJ	200-4,000 μJ
Transient Ionization Effects	Transient upset	1-100 rd(Si)/s	0.01-1 rd(Si)/s
Permanent Degradation	Ionization-induced response degradation	>10 <sup>8</sup> rd(Si)	>10 <sup>8</sup> rd(Si)
	Ionization-induced dark current increase	$10^4 - 10^6$ rd(Si)	$10^4 - 10^6$ rd(Si)
	Neutron-induced response degradation	$10^{13} - 10^{14} \text{ n/cm}^2$	$10^{13} - 10^{14} \text{ n/cm}^2$
	Neutron-induced dark current increase Electron-induced response degradation	$10^{10} - 10^{13} \text{ n/cm}^2$	$10^{10} - 10^{14} \text{ n/cm}^2$
		$10^{14} - 10^{15} \text{ e/cm}^2$	$10^{14} - 10^{15} \text{ e/cm}^2$
	Electron-induced dark current increase	$10^{11} - 10^{14} \text{ e/cm}^2$	$10^{11}$ -10 <sup>14</sup> e/cm <sup>2</sup>
nermomechanical fects	Mechanical failure	1 cal/g(Au)	l cal/g(Au)

 Table 12. Damage threshold of photodetectors

<sup>&</sup>lt;sup>62</sup> Radiation Testing of Photodetectors, Final Report Contract F29601-76-C-0034, by WH Hardwick and AH Kalma, February 1978

## **Catastrophic Damage**

This is an area in which essentially no work on photodetectors has been performed, but the similarity of photodetectors to other silicon devices can be used to estimate thresholds (ref 17,19,63-65).

For ionization-induced burnout, sensitive silicon devices are damaged at  $10^9$  rd(Si)/s, so this is taken as the threshold. For electrical pulse burnout, all silicon junction devices behave similarly, and their thresholds can be calculated by the Wunsch-Bell/Tasca model of junction burnout (ref 11, 12). In this model the thresholds depend on the semiconductor junction area (which is not necessarily the active optical area of a photodetector), allowing results for other devices to be scaled by the area ratio. The question that arises in scaling photodetector data is that their junction areas are larger than those of normal devices, and the burnout area may be less than 1/10 of the junction area assumed by the model. This would mean that the formula overestimates the threshold. To allow for this possibility, the estimates shown in table 12 are 10 percent of the scaled values so as to be pessimistic. The range of thresholds shown for any device type is caused by the range of device sizes.

However, it has been calculated (ref 66) that the fields in the depletion region of a pin photodetector will begin to collapse at dose rates of  $10^7 \text{ rd}(\text{Si})/\text{s}$ ; the radiation-induced photocurrents will tend to saturate at levels above this threshold and the recovery times will become longer with increasing dose rates.

The ionization-induced burnout mechanisms of these devices are the same as for normal silicon devices, and hardening by normal techniques (eg, current limiting) can be used on them. For all but the smallest devices, the electrical pulse burnout thresholds are high. However, the thresholds are only estimates which are uncertain, even by the low accuracy of burnout threshold standards. Therefore, the greatest need in this area is an experimental investigation of the thresholds so that the area can be properly assessed.

#### **Transient Ionization Effects**

Photodetectors are designed to be very sensitive to optical radiation. As a result, they are also very sensitive to radiation ionization, a similar phenomenon. Taking the definition of a rad and converting the energy deposited into electron-hole pair current in silicon, the theoretical current produced in devices without gain is

 $I = 6.4 \times 10^{-6} V \dot{\gamma}$ .

where

and

I = amperesV = the volume in cm<sup>3</sup>,

 $\hat{\gamma}$  is the dose rate in rd(Si)/s.

63 DNA 1420H-1 TREE Handbook vol 1 by RK Thatcher, 1971

<sup>65</sup> Gulf RT C12375, Status of Integrated Circuits, by LD Cotter and MA Donaldson, November 1975

<sup>66</sup> Gwyn, CW, Analysis of Radiation Effects in Semiconductor Junction Devices, IEEE Trans Nucl Sci NS-14, No 6, December 1967

<sup>&</sup>lt;sup>64</sup> HDL DS-94-1, Radiation Effects on Semiconductor Devices Summary Report, Components Response Information Center (DNA Data Bank), June 1974

Data on pin diodes used as ionization detectors and the (sparse) experimental information available on detectors (ref 16, 17, 28) tend to substantiate this theoretical value. For devices with gain, the gain multiplies the current determined in this manner.

The limited bandwidth of photovoltaic and transistor devices and their fast roll-off ( $\sim 100$  ns) increase the threshold. The larger-area devices are somewhat more sensitive, which is why the smaller pin diodes and avalanche photodiodes show higher thresholds than the other diode devices.

All of these transient upset thresholds are low. However, since they are directly related to the optical detection ability of the devices, little can be done to increase them except to change the material from which the detector is constructed. The smallest detector possible should be used to minimize the response, and the band-width narrowed to roll off the fast ionization response. Also, the signal-to-noise ratio should be made as high as possible by increasing the optical signal, thus requiring a larger upset pulse. This indicates that over-optimization of the receiver to reduce the noise and thus increase the S/N ratio should be avoided because it could make the system more vulnerable to upset. Obviously, shielding also decreases the vulnerability.

Steady-state noise increase is similar to transient ionization upset in that the former effect is produced by the current attributable to the latter. The difference is that steady-state effects are produced by a continually present source of ionization, such as the space-electron environment or a delayed-gamma environment. The primary effect of the steady-state ionization is to increase the noise current in the photodetector. Once the ionization ceases, the steady-state ionization effects will cease as well, so they are actually transient.

In principle, the radiation noise current may be calculated in the same manner as the transient upset signal, since the calculated radiation-induced noise current is the same type of "white" noise as the "dark" current noise. However, in these calculations the average current produced per penetrating ionization particle must be taken into account. Each type of ionization particle produces different amounts of electron-hole pairs, and, therefore, the average current per event is different for different particle fluxes. But the rms value for the radiation-induced noise current scales as the square root of the dose rate in the material, independent of the radiation particle (ref 3). The threshold is independent of system bandwidth for all but the photovoltaic types, where a maximum system bandwidth is assumed. These calculations produce thresholds which agree reasonably with those found in the few experiments that have been performed on pin photodiodes (ref 53,54,67,68).

As in the case of transient upset vulnerability, the steady-state ionization vulnerability cannot be usefully lessened in a given material by material property changes because it is related to optical sensitivity. However, switching to completely different semiconductor materials such as GaAs may raise the threshold because the generation rate in the materials may be reduced due to larger bandgaps. In fact, some of these detectors are being developed for the longer-wavelength LEDs, and these may have a higher threshold.

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<sup>&</sup>lt;sup>67</sup> IRT Report INTEL RT 0040-001, Test of MIRIS Radiometer Detectors, by AH Kalma, August 1976

<sup>&</sup>lt;sup>68</sup> IRT Report INTEL RT 0039-001, Irradiation Testing of Optics Components, by AH Kalma and RA Cesena

#### **Permanent Degradation**

Permanent ionization or displacement effects could increase the device surface leakage and thus the noise, or degrade the device gain and thus the optical response. Unfortunately, very little data exist in the area of noise increase (ref 3,53,68), and only little more in the area of responsivity degradation (ref 28,53,54), and most of these are for pin photodiodes. Estimates of the vulnerability of the other device types are made from knowledge of similar effects in normal silicon devices (ref 17,19,63,65,67). All of the diode devices except the photovoltaic cells are usually biased to depletion, which means that the response is governed by carrier sweepout and not lifetime, so they should be relatively hard to displacement damage. The photovoltaic cells should be similar in vulnerability to solar cells because both are unbiased "ON" junction devices. The phototransistors should be similar to other transistor devices. In most cases, the thresholds are only estimates with little or no experimental justification.

The dark-current changes are likely to be dependent on device geometry and construction, and it is very difficult to estimate damage thresholds. Therefore, the numbers in the table have a very large range. This is not true for electron effects on photovoltaic cells, based on data from a silicon vidicon tube (ref 69) and solar cell experiments (ref 70). An experiment examined changes in dark current in large-area pin diodes and found increases of about a factor of 3 to 5 after  $1.3 \times 10^{11}$  n/cm<sup>2</sup> (ref 53). Since large-area devices are likely to be most vulnerable, this threshold supports the estimates.

## **Thermomechanical Effects**

Since photodetectors will probably be protected from the direct X-ray beam, it is unlikely that they will suffer thermomechanical damage. The most vulnerable point of the devices will be the wire bonds; the listed thresholds are those determined for devices without wire bonds.

# RECEIVERS

Because of the very low-level signals produced in the photodetectors, amplification of these signals is necessary before any further signal processing can take place. This is usually accomplished in several stages of specially designed amplifiers. However, the first stages are most critical because most of the receiver's performance parameters are attained there. The first stages are designed to transform the nanoamperes of optically generated photocurrents in the detector to voltage levels on the order of 50 to 200 millivolts while adding as little distortion and noise as possible. Therefore, these stages are required to have high gain at low currents, low noise, and usually high frequency response. These requirements place restrictions on the semiconductors that may be employed to implement the amplifiers; therefore, some basic predictions of the radiation responses of the amplifiers are possible even though no specific design is being analyzed.

<sup>&</sup>lt;sup>70</sup> Jet Propulsion Laboratory Report 21945-6001-RV-00, Solar Cell Radiation Handbook, by JR Carter, Jr. and HY Tada, June 1973



<sup>&</sup>lt;sup>69</sup> Brucker, GJ, and Cope, AD, Radiation Sensitivity of Silicon Imaging Sensors on Missions to the Outer Planets, IEEE Trans Nucl Sci NS-19, 147, 1972

The sensitivity of the amplifiers makes them very susceptible to any radiated em energies, which means that upsets in EMP environments should occur at much lower levels unless the amplifiers are heavily shielded. Burnout is not expected to be a problem since the only electrical connections are to the power supplies and the signal output. The power supply leads are easily filtered, and the outputs should not have long cables connected to them.

In the dose-rate environment, providing high threshold levels for upsets is expected to be difficult because of the high gain requirements of the amplifiers. Radiation-generated photocurrents within the semiconductors, unless they are fully compensated, should be larger than the signal levels at moderate dose rates,  $\sim 10^5$  to  $10^6$  rd(Si)/s. However, they should not be as sensitive as the photodetectors because the semiconductors would have smaller radiation-sensitive volumes. Similarly, in the steady-state radiation environments, ionization-induced noise will decrease the S/N ratio, but this decrease is again not expected to be as large as that in the photodetectors.

Recovery of these amplifiers from the neutron flux will require a longer time because of the lower operating currents. The short-term annealing does not recover as rapidly at these levels ( $\leq 1 \ \mu A$ ) as it does in the milliampere range.

The total-dose components and those which produce displacement damage are also expected to present a problem for these amplifiers. The low quiescent operating currents required by the stages will be very sensitive to increases in leakage currents, and the high gain requirements will be sensitive to gain degradation. However, some designs for the amplifiers (transimpedance) use large amount of feedback and should, therefore, compensate for some of these effects.

Few of these devices are available from the IC manufacturers, but more should become available in the near future. One, the THEF 151 preamplifier, manufactured by Texas Instruments, has undergone limited radiation-testing (ref 3). For this device the dose-rate effects (induced noise and photocurrent) were about an order of magnitude less than that in the HP 5082-4201 photodetectors; in the total-dose and particle flux environments <5 percent gain degradation and shifts in quiescent operating voltages and currents have occurred.

#### CONNECTORS AND COUPLERS

Connectors and couplers ion various components in fiber optic systems. In all cases, this joining is from a fiber to another component (which may also be a fiber). Generally, the connections are made simply by butting the components. Tensing and other focusing schemes may be used, but they make alignment much more critical. As a result, most of the connectors are mechanical alignment devices, although index-matching fluids are sometimes used to decrease loss.

With a short optical path in the index-matching fluid, the radiation-induced absorption coefficient or luminescence would have to be extremely high to displace radiation effects in the fibers. Furthermore, it is unlikely that radiation would affect the mechanical properties of the connector before it would affect the semiconductor devices. Therefore, connectors can be considered to be unaffected by radiation.

Couplers are more complicated than connectors since they must divide or mix the light as well as mechanically join the components. Usually they make use of a short length of optically transparent material, which would likely be one from which fibers are made.

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Degradation	Specific	
Level	Effect	Coupler
Transient Ionization Effort	Transient upset signal	10 <sup>6</sup> - 10 <sup>9</sup> rd(Si)/s
Enects	Transient signal loss (immediate)	10 <sup>11</sup> - 10 <sup>12</sup> rd(Si)/s
	Transient signal loss (1 ms)	10 <sup>11</sup> - 10 <sup>12</sup> rd(Si)/s
	Transient signal loss (1 s)	10 <sup>11</sup> - 10 <sup>14</sup> rd(Si)/s
Permanent Degradation	Ionization-induced signal loss	10 <sup>5</sup> · 10 <sup>8</sup> rd(Si)/s
Thermomechanical Effects	Mechanical failure	10 cal/g(Au)

#### Table 13. Damage threshold of couplers

Radiation could affect couplers by producing absorption or luminescence in the optical material. Because of the relatively short length of a coupler compared to a fiber, it is more likely that the fiber would set the system vulnerability level. However, couplers which must be radiation-tolerant must not be constructed of radiation-soft glass or other radiation-soft material, since they are long enough that the additional absorption produced by radiation would adversely affect system hardness. Most couplers currently being built have not been designed with radiation hardness in mind: therefore, it is possible that some would not have sufficient hardness. Although experimental test data are not available, it is probably not necessary to radiation-test these components, but merely to exercise reasonable material choice in their design and construction. Choice of the proper materials can be made using the information available from fiber tests, unless some entirely new material (other than glass or plastic) is used.

Thresholds were estimated from this information, and the results are listed in table 13. Coupler length of 2 cm was assumed. Radiation-soft couplers are assumed to be made from doped-silicate glass or doped silica, and radiation-hard couplers are assumed to be made from the appropriate glass or plastic material.

## SYSTEM TRADEOFF CONSIDERATIONS

#### **INTRODUCTION**

In earlier sections of this report, the radiation vulnerabilities of fiber optic components were presented. This section considers the system design tradeoffs necessary to maintain system performance during and/or after exposure to a radiation environment when the above mentioned components are used. For purposes of this discussion, the system considered will consist of general-purpose black boxes which transmit information via fiber optic cables. The communication system, whether it is analog or digital, is designed to transfer information at some rate from one black box to another. A consideration for the system designer is whether the disruption of the information transfer can be tolerated, and, if so, how long it may exist before becoming detrimental to the system. Once these questions are answered, the designer should determine which of the component vulnerabilities will produce disruption of the information transfer. The vulnerabilities should then be categorized as to whether they produce (1) catastrophic failures, (2) permanent degradation, or (3) transient upsets in the system, as discussed in previous sections.

## **CATASTROPHIC FAILURES**

For components having vulnerabilities which will produce catastrophic failures, radiation protection must be incorporated into the design. Various methods of providing radiation protection are available and are similar to those required for hardening of electronic circuits. These include shielding, circumvention, EMP protection, photocurrent compensation and/or limiting.

#### **Light Sources**

As discussed previously, burnout of light sources is possible at some system threat levels, and shielding may be a solution to this problem. This is possible because the mounting for these components provides two functions — heat sinking and optical alignment. With proper design and choice of materials for the mountings, they may provide radiation shielding. However, to provide enough shielding to reduce the threat level to below that which will produce catastrophic failures may result in size and weight increases. The choice of materials is critical because they must provide attenuation of the radiation and good thermal conductivity.

Additional hardening may be added at the circuit level by limiting the maximum current flow through the device to below that which will produce an instantaneous power dissipation that would be required to reach the burnout threshold. This may be accomplished by inserting a resistor in series with the light source. However, this resistor and the device junction capacitance will now provide an RC rolloff which may be below the system's required bandwidth. To compensate for the rolloff, the resistor may be bypassed by a capacitor, the size of which must be limited to a value that will not store enough energy to cause burnout in the light source.

For circuits in which the above current-limiting technique cannot be employed because of system constraints such as bandwidth, circumvention may be used to interrupt the power source at the circuits. This will result in a loss of data transfer during the circumvention period, but will prevent burnout of the light source and its drive circuitry.

## **Light Detectors**

For the purposes of this discussion, it is assumed that the types of semiconductor light detectors most likely to be used in military systems are the pin and avalanche photodiodes. These devices are similar in circuit operation in that they both are reverse-biased into depletion. Because they are light detectors they also respond to incident ionizing radiation; therefore, photocurrent protection is essential. Here again, because these devices are usually mounted in connectors that provide optical alignment with the fibers, it is possible to select materials for their construction that will provide radiation shielding. The obvious weight and size tradeoffs are the governing factors.

Since photocurrent generation in a device is directly proportional to the volume of the active region in the detector, it is beneficial to reduce the volume as much as possible. This would indicate that detectors having the smallest active area should be used in critical systems. However, the design tradeoff here is that of collecting enough light signal exiting the fiber optic cable. Since the light exits the cable at its numerical aperture, the design should place the end of the fiber optic cable as close as possible to the detector.

Also, reduction of the area of the junction implies that the burnout threshold is lower (ref 11,12). It is then necessary to maximize the area/volume ratio. To limit the current flow through the detector to prevent burnout, a resistor may be added between the bias supply and the detector; however, this resistor now would produce an additional noise source at the input to the amplifier. It should, therefore, be bypassed by a capacitor whose size must be limited.

Circumvention may also be considered for the light detector bias circuitry to provide radiation protection. But because the bias supply may be a relatively high voltage supply (>50V) derived from a lower one (ie, a dc-to-dc converter), the time required to return to the designed detector bias voltage would be greater than the circumvention time; thus, signal outage time may be increased. Some state-of-the-art pin diodes may be operated from the 5-V logic supply bus, which makes circumvention very practical for these devices.

## PERMANENT DEGRADATION

Some of the radiation damage produced within the fiber optic components results in permanent degradation. Some effects may be annealed out by baking the components at elevated temperatures for relatively long time periods (minutes to hours to days), but at normal operating temperatures the parameter degradation may be considered permanent. Parameter degradations may result in a decrease in the electro-optical conversion efficiencies of semiconductor components or in increased attenuation coefficients for fiber optic cables, causing a decrease in the quality of information transfer between the fiber optics system black boxes.

#### Light Sources

The major concern for the designer when using semiconductor light sources in radiation environments is how to compensate for the decrease in light power output of these devices. One method is to use brute force and increase the operating current through them to allow for the decrease in optical power output. Although this will solve the power problem, the method has shortcomings; it decreases the operating lifetime of the source. Therefore, when the designer uses this method he must consider this consequence as well as the increased electrical power and heat-sinking requirements.

Another method is to use an active feedback circuit to adjust the drive current through the source. This method has the feature of not increasing the current through the device until degradation has occurred. It has the disadvantage that an optical detector must be placed in such a position that it samples the light output of the source. It is for this reason that the method is usually applied only to edge-emitter sources so that the detector and its amplifier circuitry may be placed behind the chip in the same package. This increases the cost of the source and the complexity of the light source drive circuitry.

A unique problem exists when using laser diodes in a radiation environment; the threshold current (the minimum current at which lasing action will begin) will increase with increasing radiation. This will result in two detrimental effects on the system — reduction of dynamic range and slower turn-on. For analog systems in which the signal's dynamic range is important, the maximum signal swing is limited by the difference between the quiescent current-biasing point and the threshold current. Therefore, when the threshold current increases, signal distortion will occur at lower signal levels. For those systems in which the high-frequency response of the laser diode is important, the quiescent operating current is selected to be just above the threshold current level, to minimize the turn-on time. If the threshold current increases above that of the quiescent current due to radiation exposure, a decrease in the high-frequency response is inevitable. If the designer tries to compensate for this shift by increasing the quiescent bias current, he must consider that the on-to-off light power level ratio will be reduced, and that the power dissipation in the laser diode will increase and will result in a shorter operating lifetime.

### Cables

To compensate for the increase in fiber absorption caused by radiation, the designer has four areas in which he may work. The first is the choice of materials from which the fiber optic cable is manufactured, as discussed earlier, along with their relative radiation responses and optical properties. Second, the relative power output of the light source may be increased to provide an adequate design margin after radiation. The third method is to provide radiation shielding, which may not be a solution for some systems because of cable size and weight increases. Finally, the system architecture may be designed to allow for system outage times.

The major concerns for the designer when using semiconductor photodetectors in radiation environments are dark current increases and reduced device gain, or conversion efficiency and noise generation.

Dark current increases are particularly bothersome to systems in which de signal response is required because this current is summed with the signal current at the detector amplifier output. Under normal design practices, this current is at least an order of magnitude below the minimum detectable light signal; however, it is possible that, at some system threat levels, the dark current could increase by two to three orders of magnitude. This could easily cause a dc-responding amplifier to be forced out of the linear region into saturation, resulting in loss of signal. Since this current originates in the surface areas and depletion regions of the semiconductor junctions, it would be beneficial to select devices that have small active areas, as discussed earlier. To compensate for the increase in current into the summing point at the amplifier input, a matched diode that is not exposed to the input optical signal may be added to provide an alternate current path around the summing point.

In systems in which adequate S/N ratios exist automatic gain control (AGC) may be designed into the photodetector amplifier to compensate for the reduced gain or conversion efficiency in the detector. This will result in an increase in the complexity of the amplifier design and in the analysis of amplifier response to transient radiation exposure.

Shielding may also be employed to provide radiation hardening to protect the detector from dark-current increases.

## **TRANSIENT EFFECTS**

Transient radiation effects on the fiber optic components produce an addition of false signals into the data and/or temporary loss of signal. The designer must determine if these effects will produce detrimental system response, and design against them if they cannot be tolerated.

#### **Light Sources**

In systems in which data transmission is required during the radiation pulse, particular care must be taken to compensate for photocurrents generated within the source's modulating circuitry. The photocurrents must be limited to a level that will result in a current through the source that will produce a minimum detectable optical signal from the source. This may require matching photocurrent responses in devices, which becomes a very difficult procurement problem, or providing radiation shielding to the entire modulating circuitry.

For systems in which data interruption may occur but no false signals may be produced, circumvention is a possibility. As long as the system does not accept information during this time, it does not matter if the source is momentarily turned on or off.

#### Cables

Transient ionization effects in fiber optic cables produce two phenomena that are of concern to the designer - luminescence and increase in attenuation. Their mechanisms and their effects on the system were discussed earlier. This section concerns system design alternatives that may be used to reduce or eliminate the unwanted responses from the system.

The luminescence problem exists in only those systems that require data transmission during the radiation, and may be reduced by adding optical bandpass filters between the end of the cable and the photodetector. This will result in increased attenuation of optical signals outside the transmitted bandwidth, with little attenuation of the data signal. Care should be taken when selecting materials from which to construct the filters to choose materials that will not luminesce. Adding the filters at the detector will increase the complexity of the connector design and will necessarily increase the distance between the end of the fiber optic cable and the detector. This will result in further spreading of the exiting light, which may require a larger-area detector or lensing to collect all the available light signal. If pigtailed detectors are used, the filters have to be added at the connector that mates the end of the pigtail with the end of the fiber. This will increase the connector losses for most connectors significantly, and will virtually eliminate all low-loss connectors (<1 dB).

Transient absorption increases in fiber optic cables present a problem to the designer because they can require relatively long periods in which to recover, resulting in significant periods of signal outage which some systems cannot tolerate. This problem can be counteracted by increasing the power output of the light source and, for systems with large S/Nratios, using AGC circuits in the photodetector amplifier circuitry. The consequences of both these actions have been discussed previously. The reduction in outage time may not be significant for large induced transient absorption (much greater than the design margin). With some of the newer fiber optic cables this problem has been greatly reduced and the choice of one of these cables may eliminate the problem. For ground-based systems, the cables may be buried in the earth to provide the required shielding.

## Detectors

For systems that must operate through transient radiation exposures, a significant problem exists with the photodetectors. The photocurrents generated within the detectors may be several orders of magnitude larger than the optical signals; therefore, some method of preventing these currents from entering the detector amplifier circuitry must be found. One design solution is to use radiation shielding to protect the detector, the consequences of which have been discussed earlier. Another method is to add another matched diode that is not exposed to the input optical signal and which will provide an equal but opposite photocurrent out of the summing point. The added diode should not be confused with the "guard-ring" diode contained in some detectors. Adding this diode may create a problem in systems in which small S/N ratios are anticipated, since its noise will add to the noise of the optical detector and reduce the S/N ratio by 3 dB.

Exposure of the photodetector to steady-state ionizing radiation will produce a noise signal that will degrade the system S/N ratio. To provide rejection of out-of-band noise, the designer should limit the bandpass of the detector amplifier circuitry to the minimum required for data transfer. This will complicate the circuit design of the amplifier and may put additional constraints on its open-loop gain and frequency response. In systems in which large bandwidths are required, it will be necessary to provide shielding for the photodetector or to increase the optical signal level at the input to the detector. Earlierdiscussions describe the consequences of both these actions.

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# SUMMARY EVALUATION OF THE APPLICABILITY OF FIBER OPTICS TO GENERIC MILITARY SYSTEMS

The vulnerabilities of fiber optic systems are compared to environment levels for various system types. For all levels, 120 dB attenuation of the free-field EMP is assumed. These generic examples provide gross guidelines for applicability to military systems.

# **RE-ENTRY VEHICLE SYSTEMS**

For cables installed inside the heat shield but outside the electronic component X-ray shield the ionization dose delivered to a fiber optic cable is such that none of the systems discussed in this report will recover operation within 10 minutes of exposure. Even for the doses assumed to be within the X-ray shield, the links will require 1 second or longer to recover to operability. No fiber optic system examined in this report appears to be readily applicable to a re-entry vehicle system.

## SATELLITE SYSTEMS

EMP fields are of relatively small magnitude for satellites, thus satellite data links are not expected to be significantly vulnerable. For satellites hardened to JCS levels, fiber optic systems have limited applicability, and then only if shielding is provided. The amount of shielding required is such that there is no advantage over a hard-wired link from a radiation effects point of view. Fiber optic systems are applicable to hardened satellites at levels near  $10^{-3}$  JCS, and on unhardened satellites provided the photodetector is adequately protected against noise from electrons in the belt. The amount of protection required is orbit and scenario-specific.

### **STRATEGIC SYSTEMS**

The large prompt dose absorbed by strategic missile components when exposed to the in-flight radiation environment makes all the fiber optic systems considered here inapplicable without considerable shielding. The amount of shielding is such that fiber optic systems have no advantage in weight or size compared to a hard-wired link.

None of the links will operate through the example environment without further shielding. The links will recover from both the flight threat example and the ground example in 10 ms or less. For an aircraft which may be exposed to the EMP threat without the direct radiation environment, only the 1, 10, and 100-Mbit point-to-point links will meet the operate-through requirement attenuated by 120 dB for the EMP threat.

For ground facilities, all the fiber optic systems considered here will recover from exposure to the example threat within 1 ms of exposure if the entire system is protected by two metres or more of earth. Because of the EMP environment, the operate-through requirement can be met for less than 100 m of cable by burial of the electronics at a depth of five metres or greater.

In the 1 to 10-ms time regime, the delayed components of the radiation environment cause inoperability of the link. For scenarios for which there is negligible (less than  $10^{-3}$  rd) ionizing radiation deposited in aircraft components, the 1, 10, and 100-Mbit point-to-point lengths will operate through exposure of the aircraft to the EMP threat, assuming 120 dB attenuation of the free-field threat at the component level. It is anticipated that the other types of links will upset but will return to operation after a few hundred nanoseconds

of upset by the EMP pulse. It is for those scenarios in which the TRE environment has been reduced, either by distance or by shielding, that firm conclusions about the applicability of a fiber optic data link to a specific aircraft system must be based on an understanding of the details of the system geometry and shielding. The EMP environment at the susceptible component can easily vary over several orders of magnitude for slight changes in the electronic enclosure configuration or power supply line shielding. Hence, general conclusions about the applicability of fiber optics to aircraft systems where the TRE levels have been reduced cannot be made.

## TACTICAL SYSTEMS

The range of tactical scenarios is such that it is impossible to summarize in a brief document the applicability of fiber optic systems to a generic tactical situation. However as an example, none of the links will meet an operate-through requirement even 10 km from a 1-kt surface burst. All the links will recover 1 ms after exposure to a 1-kt surface burst at a 0.5-km slant range. At the 8-psi level (0.03 km from a 1-kt surface burst), only the 100-m, 1-Mbit link will recover in 1 ms. The remaining links require 10 seconds or longer to recover. Fiber optic data links could be considered functional in tactical situations, but the degree depends upon both the application and the anticipated scenario.

## **RECOMMENDATIONS.**

It is recommended that fiber optics be considered in the design of any system whose cable runs are greater than ten metres and which must adhere to military system EMP requirements.

It is recommended that fiber optics be considered in the design of any system whose recovery times are greater than millisecond, and whose dose rate and total dose deliverance are within the budget link design margin.

It is also recommended that fiber optics be considered in the design of any system whose operation is dependent upon the survivability of its personnel, where personnel nuclear survivability requirements are the limiting factor in the system's operation.

Additional data must be obtained to ascertain the displacement damage-producing effects on fiber optic systems.

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#### REFERENCES

- 1. Vance, EF, Coupling to Shielded Cables, Wiley-Interscience, 1978
- 2. AFAL TR-45, Opto-electronics Aspects of Avionic System II, by JR Baird, May 1975
- 3. Hardwick, WH, and Kalma, AH. Effects of low-Dose-Rate Radiation on Opto-Electronic Components and the Consequences on Fiber Optic Data Link Performance, IEEE Trans Nucl Sci NS-26, No 6, December 1979
- 4. ERADCOM TR-28-13, Nuclear Radiation Vulnerability of Proposed Army Fiber Optics Communications System, by S Kronenberg, et al, 1978
- 5. Arimura, I, and Colwell, R, Radiation Degradation and Recovery in Long Fiber Optic Data Links, presented at 1978 Symposium on Fiber Optics in Nuclear Environment, March 25-27, 1980
- 6. Sigel, GH, Jr, et al, Radiation Response of Large-Core Polymer-Clad Silica Optical Fibers, IEEE Trans Nucl Sci NS-26, No 6, December 1979
- 7. NOSC Report TR-469, EMP Hardening of Airborne Systems through Electro-optical Techniques: Design Guideline, by R Greenwell, 15 December 1979
- 8. IRT Report 4521-002, The ABCs of Radiation Hardening, April 1976
- 9. IRT Final Report for AFWL, Fiber Optic Radiation Test and Evaluation Program, Kirtland AFB, December 1978
- 10. Motorola Application Note AN-707, Noise Immunity Comparison of CMOS vs Popular Bipolar Logic Families, by AA Allen, 1973
- Wunsch, DC, and Bell, RR, Determination of Threshold Failure Levels of Semiconductor Diodes and Transistors Due to Pulse Voltage, IEEE Trans Nucl Sci NS-17, 364, 1970
- 12. Tasca, DN, Pulse Power Failure Modes in Semiconductors, IEEE Trans Nucl Sci NS-17, 364, 1970
- 13. AFWL TR-74-302. An Assessment of Fiber Optics Technology for Satellite Hardening, by AH Kalma, 1975
- 14. Kalma, AH, and Fisher, CJ, Electrical Pulse Burnout Testing of Light-Emitting Diodes, IEEE Trans Nucl Sci NS-22, 2510, 1975
- 15. O.Brien, JT, et al, Recent Developments in Injection Lasers, presented at Conference on Laser and Electro-Optical Systems, San Diego, California, 25-27 May, 1976
- AFWL TR-68-31, vol 2, Radiation Effects on GaAs Devices and Schottky Diodes, by HD Southward and RH Schnurr, August 1968
- 17. Northrop Report DN A3776F, Study of Radiation Effects in Bulk CMOS Microcircuits, IIL/LSI Logic Cells, and Optic Couplers, by JP Raymond, et al, 1975

- 18. Barnes, CE, Development of Efficient Radiation-Insensitive GaA:Zn LEDs, IEEE Trans Nucl Sci NS-24, No. 6, December 1977
- 19. AFAL TR-74-61 Survey of Vulnerability of Contemporary Semiconductor Components to Nuclear Radiation, by RP Donovan, et al, 1974
- 20. Epstein, AS, et al, Gamma Irradiation and Annealing Effects in Nitrogen-Doped GaAsP Green and Yellow Light-Emitting Diodes, IEEE Trans Nucl Sci NS-19, 386, 1972
- 21. Barnes, CE, Effects of <sup>60</sup>Co Gamma Irradiation on Epitaxial GaAs Laser Diodes, Phys Rev B1, 4735, 1970
- 22. Schnurr, RH, and Southward, HD, Radiation Effects on GaAs Devices, IEEE Trans Nucl Sci NS-15, 306, 1968
- 23. Soda, KJ, et al. The Effect of Gamma Irradiation on Optical Isolators, IEEE Trans Nucl Sci NS-22, 2475, 1975
- 24. Polimadei, RA, et al, Performance of GaA1As Light-Emitting Diodes in Radiation Environments, IEEE Trans Nucl Sci NS-21, 96, 1974
- 25. Share, S et al, Radiation Effects and Hardening on Compensated GaAs Light-Emitting Diodes, IEEE Trans Nucl Sci NS-20, 256, 1973
- 26. Epstein, AS, et al, Radiation Effects on Active and Passive Optical Components, presented at DoD/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- 27. Schroeder, JD, et al. Radiation-Damage-Induced Time Delay in GaAs Lasers, IEEE Trans Nucl Sci NS-19, 391, 1972
- 28. Epstein, AS, and Trimmer, PA. Radiation Damage and Annealing Effects on Photon-Coupled Isolators, IEEE Trans Nucl Sci NS-19, 391, 1972
- 29. Barnes, CF, Radiation Effects in Electroluminescent Diodes, IEEE Trans Nucl Sci NS-18, 322, 1971
- 30. Noel, BW, et al, Neutron Irradiation Effects on Diffused GaAs Laser Diodes, IEEE Trans Nucl Sci NS-18, 378, 1971
- 31. Barnes, CE, Neutron Damage in Epitaxial GaAs Laser Diodes, J Appl Phys 42, 1941, 1971
- 32. Schade, H, et al, Defect Centers in GaAsP Electroluminescent Diodes Due to High-Energy Electron Irradiation, J Appl Phys 41, 3783, 1970
- 33. Barnes, CE, Neutron Damage in GaAs Laser Diodes at and above Laser Threshold, IEEE Trans Nucl Sci NS-19, 382, 1972
- 34. Epstein, AS, et al, Effect of Neutron Irradiation on GaAsP Electroluminescent Diodes, Appl Phys Lett 23, 472, 1973
- 35. Stanley, AG, Comparison of Light-Emitting Diodes in a Space Radiation Environment, IEEE Trans Nucl Sci NS-17, 239, 1970

- 36. Sandia Report SC-RR-710686 Radiation Effects in Light-Emitting Diodes and Laser Diodes, by CE Barnes, 1972
- 37. Barnes, CE, Neutron Damage in GaP Light-Emitting Diodes, Appl Phys Lett 20, 110, 1972
- Compton, DMJ, and Cesena, RA, Michanism of Radiation Effects on Lasers, IEEE Trans Nucl Sci NS-14, 55, 1967
- 39. Arnold, GW, Luminescence in Intrinsic and Annealed Electron-Irradiated GaAs:Cd, Phys Rev 183, 777, 1969
- 40. Barnes, CE, Appl Phys Lett 20, 86, 1972
- 41. Barnes, CE, Tunneling in SiC Electroluminescent Diodes, J Appl Phys 45, 193, 1974
- 42. Barnes, CE, Increased Radiation Hardness of GaAs Laser Diodes at High Current Densities, J Appl Phys 45, 3485, 1974
- 43. Evans, BD, and Sigel, GH, Jr, Radiation-Resistant Fiber Optic Materials and Waveguides, IEEE Trans Nucl Sci NS-22, 2462, 1975
- 44. Mattern, PL, et al, Effects of Radiation on Absorption and Luminescence of Fiber Optic Waveguides and Materials, IEEE Trans Nucl Sci NS-21, 81, 1974
- 45. Evans, BD, and Sigel, GH, Jr, Permanent and Transient Radiation-Induced Losses in Optical Fibers, IEEE Trans Nucl Sci NS-21, 113, 1974
- 46. NRL Memorandum Report 2704, Radiation Effects in Fiber Optic Waveguides, by GH Sigel, Jr, 1973
- 47. Mattern, PL. Radiation-Induced Absorption and Luminescence in Glass and Plastic Optical Waveguides, presented at DoD/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- 48. NRL Memorandum Report 2934, Radiation Effects in Fiber Optic Waveguides, by GH Sigel, Jr, et al, 1974
- 49. AFCRL TR-74-0012, Transient Radiation Effects Tests of a Corning Radiation-Resistant Optical Fiber, by JA Wall, 1975
- 50. AFCRL TR-75-0190, Radiation Effects on Fiber Optics, by JA Wall and JF Bryant, 1975
- Mattern, PL, et al, Effects of Radiation on Optical Fibers and Materials, IEEE Trans Nucl Sci NS-22, 2468, 1975
- 52. Sandia Report SAND76-8056, A Summary of Radiation-Induced Transient Absorption and Recovery in Fiber Optic Waveguides, by CD Skoog, 1976
- 53. IRT Report INTEL RT 8111-089, Fiber Optics Test Program for Satellite Applications, by AH Kalma and WH Hardwick
- 54. Kalma, AH, and Hardwick, WH, Radiation Testing of a Fiber Optics Data Link, IEEE Trans Nucl Sci NS-23, 1769, 1976

- 55. Maurer, RD, et al, Effects of Neutron and Gamma Radiation on Glass Optical Waveguides, Appl Opt 12, 2024, 1973
- 56. Sigel, GH, Jr, and Evans, BD, Effects of Ionizing Radiation on Transmission of Optical Fibers, Appl Phys Lett 24, 410, 1974
- 57. Acharekar, MA, Multimode Radiation-Resistant Glass Fiber for Data Transmission Applications, presented at Dod/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- 58. Sigel, GH, Jr, Radiation Damage in Fiber Optics, presented at DoD/Industry-Wide Integrated Optics and Fiber Optics Communications Conference, NELC, San Diego, California, 1974
- 59. Friebele, EJ, et al, Radiation Protection of Fiber Optics Materials: Effects of Oxidation and Reduction, Appl Phys Lett 24, 412, 1974
- 60. Friebele, EJ, et al, Defect Centers in a Germanium-Doped Silica-Core Optical Fiber, J Appl Phys 45, 3424, 1974
- 61. Baba, A, et al. Effect of Simulated Nuclear Thermal Pulses on Fiber Optic Cables, IEEE Trans Nucl Sci NS-26, No 6, December 1979
- 62. Radiation Testing of Photodetectors, Final Report, Contract F29601-76-C-0034, by WH Hardwick and AH Kalma, February 1978
- 63. DNA 1420H-1 TREE Handbook, vol 1, by RK Thatcher, 1971
- 64. HDL DS-94-1, Radiation Effects on Semiconductor Devices Summary Report, Components Response Information Center (DNA Data Bank), June 1974
- 65. Gulf RT C12375, Status of Integrated Circuits, by LD Cotter and MA Donaldson, November 1975
- 66. Gwyn, CW, Analysis of Radiation Effects in Semiconductor Junction Devices, IEEE Trans Nucl Sci NS-14, No 6, December 1967
- 67. IRT Report INTEL RT 0040-001, by AH Kalma, August 1976
- 68. IRT Report INTEL RT 0039-001, Irradiation Testing of Optics Components, by AH Kalma and RA Cesena
- 69. Brucker, GJ, and Cope, AD, Radiation Sensitivity of Silicon Imaging Sensors on Missions to the Outer Planets, IEEE Trans Nucl Sci NS-19, 147, 1972
- 70. Jet Propulsion Laboratory Report 21945-6001-RV-00, Solar Cell Radiation Handbook, by JR Carter, Jr, and HY Tada, June 1973

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