

AD-A080 491

MASSACHUSETTS INST OF TECH LEXINGTON LINCOLN LAB

F/6 9/5

CHARACTERIZATION OF TRANSCONDUCTANCE MULTIPLIER WEIGHT CIRCUITS--ETC(U)

JUN 79 B C LEVENS

F19628-78-C-0002

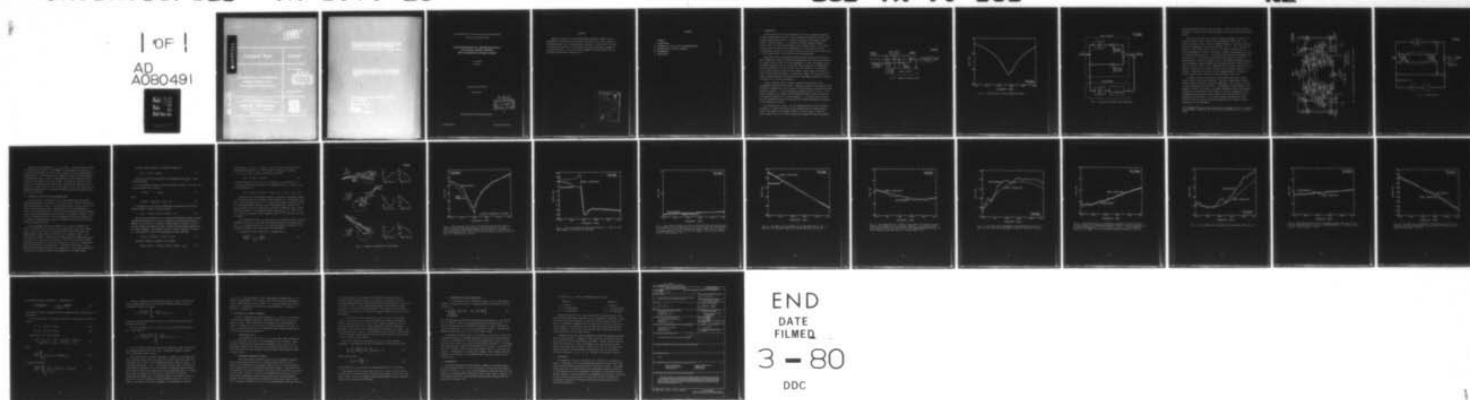
UNCLASSIFIED

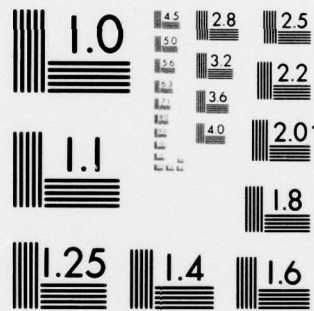
TN-1979-23

ESD-TR-79-161

NL

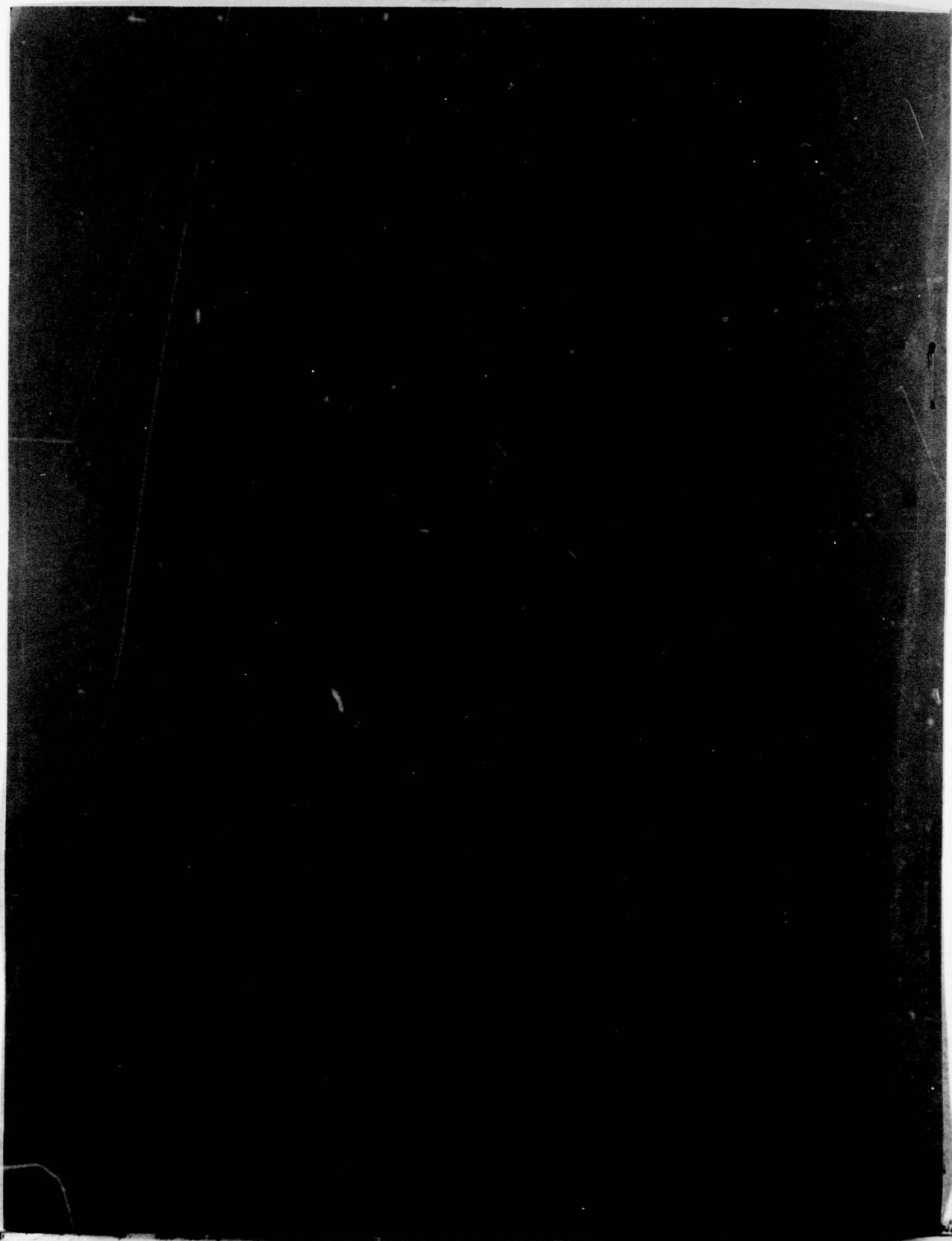
1 OF 1  
AD  
A080491





MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A

ADA080491





MASSACHUSETTS INSTITUTE OF TECHNOLOGY  
LINCOLN LABORATORY

CHARACTERIZATION OF TRANSCONDUCTANCE  
MULTIPLIER WEIGHT CIRCUITS  
FOR AN ADAPTIVE ANTENNA SYSTEM

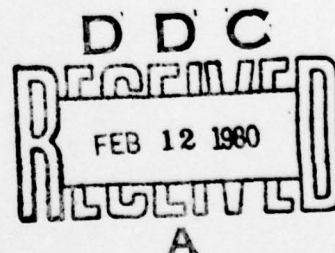
B. C. LEVENS

Group 68

*for 1473  
in  
back*

TECHNICAL NOTE 1979-23

22 JUNE 1979



Approved for public release; distribution unlimited.

LEXINGTON

MASSACHUSETTS

# ABSTRACT

Extensive testing of the transconductance multiplier weight circuits designed for the antenna nulling project indicates that a simple time delay feedthrough model fits the measured data very well. Properties of this model are derived and applied to the problem of determining the model parameters from measured data. Plots of actual measurements and model predictions are provided to demonstrate goodness of fit.

Accession For	
NTIS G1241	<input checked="checked" type="checkbox"/>
DDC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By _____	
Distribution/	
Availability Codes	
Dist	Avalland/or special
A	

## CONTENTS

ABSTRACT	iii
I. INTRODUCTION	1
II. PROPERTIES OF TIME DELAYED FEEDTHROUGH MODEL	8
III. DETERMINATION OF MODEL PARAMETERS	24
IV. MEASUREMENTS	26
V. CONCLUSIONS	27

## I. INTRODUCTION

Recently, there has been much interest in the use of Adaptive Antennas in Space Communications systems. These antennas are expected to cancel interference signals automatically. The basic structure of an Adaptive Antenna system is shown in Figure 1. It consists of several antenna elements, a weight circuit for each antenna element, and a control circuit. The control circuit senses the direction and magnitude of the interference and sets the amplitude and phase of the weight circuits in such a way as to form a null in the antenna pattern in the direction of the interference.

Experience with an experimental Adaptive Antenna system has shown that when attempting to obtain interference cancellation over a non-zero bandwidth the weight circuits often limit the performance. In fact, the weight circuit performance appears to be one of the most critical features in determining how much cancellation can be obtained. For this reason, it is desirable to have a simple analytical model for the weight circuits which can be used to predict cancellation performance. Having such a model, one could specify in terms of model parameters the weight circuit requirements necessary to achieve a desired level of system performance.

Several different types of weight circuits were evaluated in the experimental Adaptive Antenna system. Though they differ in configuration and performance significantly, the circuits all have roughly the same characteristic frequency response. A typical frequency response is shown in Fig. 2. This figure represents the response obtained when the weight circuit is adjusted for maximum attenuation at a single frequency. It is clear from the figure that the achievable attenuation averaged over a band of frequencies decreases as the bandwidth increases.

The resemblance of Figure 2 to such familiar patterns as the frequency response of a single-zero digital filter suggests that the simple model of Figure 3 may apply for the weight circuits. In this figure we model the actual weight circuit as an ideal weight with an additive term of unweighted,



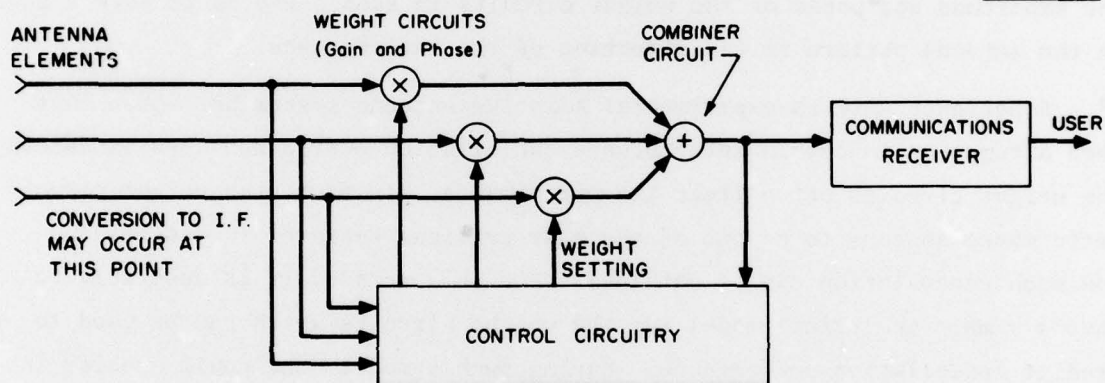


Fig. 1. Adaptive antenna system.

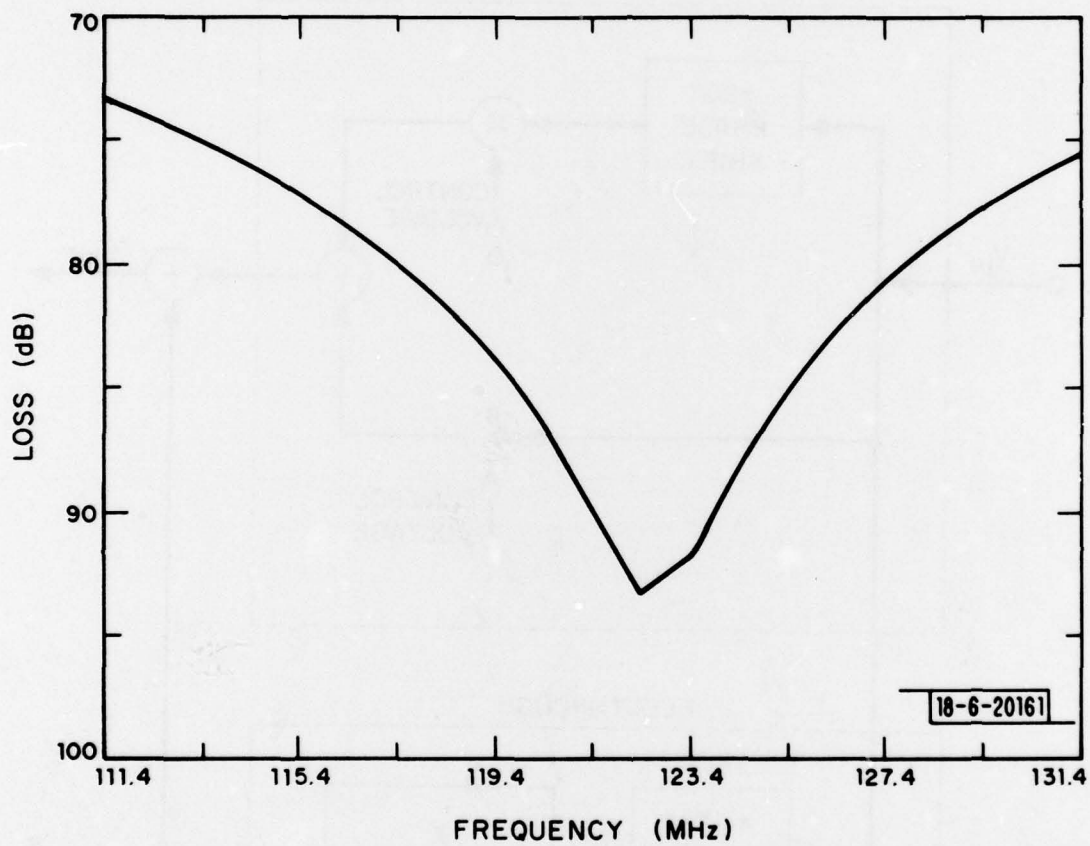


Fig. 2. Typical weight circuit frequency response.

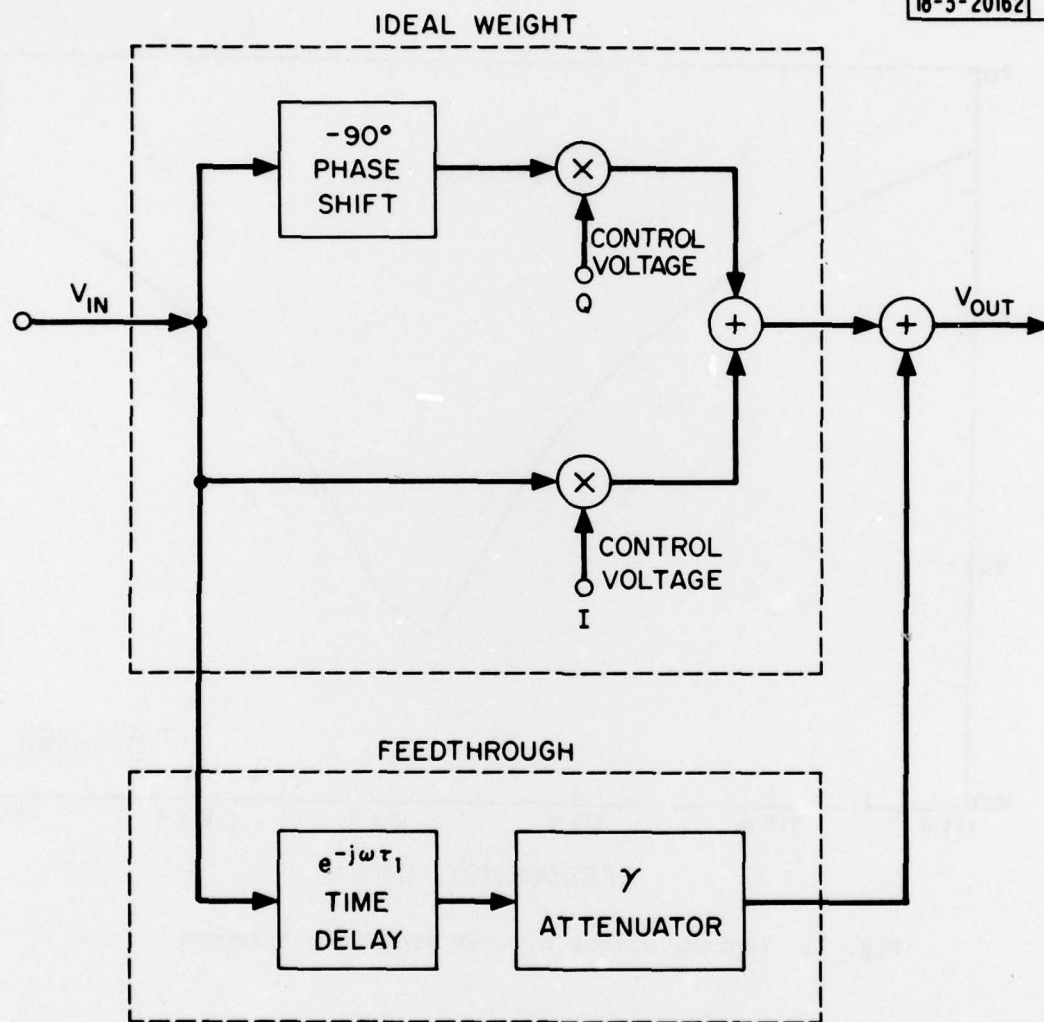


Fig. 3. Weight circuit model with feedthrough.

but delayed feedthrough from input to output. Despite this model's apparent simplicity it has been found to be applicable to a variety of different weight circuits.

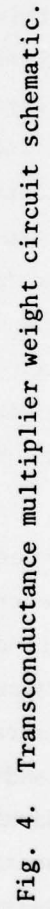
This report addresses the problem of fitting the model of Figure 3 to a particular weight circuit design--the transconductance multiplier. The general structure of this circuit is shown in Figure 4. Note that the weight setting is determined by two control voltages  $V_I$  and  $V_Q$ . For a complete description of this circuit see TN 1979-11\*. In order to confirm that time delayed feedthrough is indeed a valid model, extensive measurements were made on individual weight circuits. Calculations were then made to determine what values of  $\gamma$  and  $\tau$  (from Figure 3) give the best overall fit of model to measurement for a particular weight circuit. The measured frequency response and the frequency response predicted by the model were then plotted on the same set of axes so that a visual comparison could be made. These plots were made for a wide variety of weight settings over a frequency range of 121.4  $\pm$  10 MHz. Inspection of the plots (see for example Figures 7 through 16) reveals that the model predicts the observed behavior very well.

It should be borne in mind that a more complete model must include several effects other than feedthrough. These are illustrated in Figure 5. Several features of Figure 5 should be emphasized. First, there is an over-all delay associated with the device. The mismatch of this parameter from channel to channel (see Figure 1) is another potential source of performance degradation in the antenna system; however, it is not as important as the feedthrough because measures can be taken to trim out variations in this parameter (see TN 1979-11). The over-all delay is mentioned here only because it must be subtracted from the measured data in order to obtain a measurement of the feedthrough.

---

\*J. N. Wright, "Transconductance Multipliers for Weight Circuits in an Adaptive Nulling System," Technical Note 1979-11, Lincoln Laboratory, M.I.T. (1 June 1979)





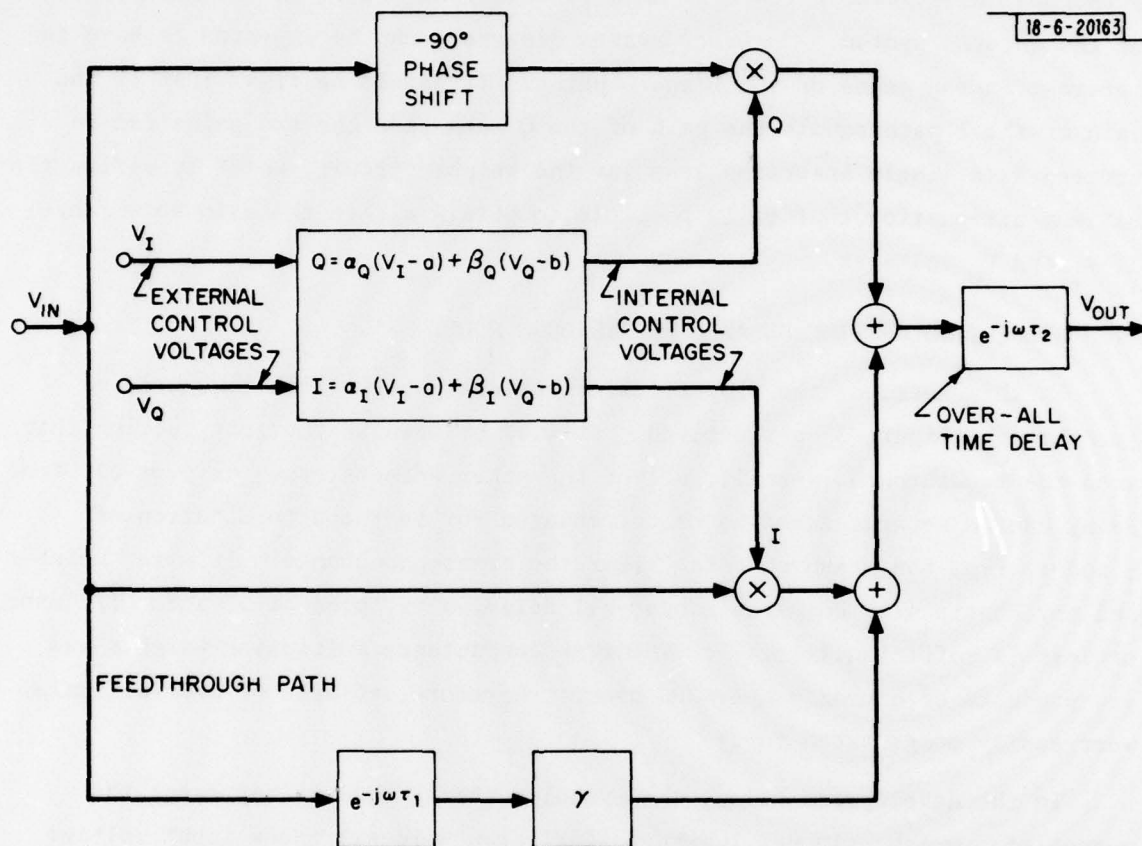


Fig. 5. Complete model.

Second, are the parameters  $\alpha_I$ ,  $\alpha_Q$ ,  $\beta_I$ , and  $\beta_Q$ . These have been included to indicate that the gains of the inphase (I) and quadrature (Q) paths are not precisely equal and that the  $V_I$  and  $V_Q$  voltages do not represent perfectly orthogonal axes. These features have virtually no effect on the performance of the antenna system. However, system designers can be expected to have an interest in the gains of the I and Q paths. It should be clear that if the gain of the I path equals the gain of the Q path then the two gains can be treated as a single insertion loss for the weight circuit, which specifies the minimum attenuation that it is possible to obtain within the allowable range of  $V_I$  and  $V_Q$  voltages.

## II. PROPERTIES OF TIME DELAYED FEEDTHROUGH MODEL

In this section, the properties of an ideal weight with time delayed feedthrough (Figure 3) are derived. It will be seen in the next section that when the feedthrough is small, all of the other effects, such as over-all time delay can be separated and hence compensated for in the determination of  $\gamma$  and  $\tau_1$  from the measured data. (For the transconductance multiplier weights and any others with comparable over-all delay, a  $\gamma\tau_1$  product of about 1/2 nsec. or less is sufficiently small. The transconductance multiplier weights had  $\gamma\tau_1$  products of approximately 250 psec.; therefore, effects of over-all delay were easily compensated for.)

In the development below, the symbol  $V_o(t)$  is used to represent the output of a weight circuit (whether ideal or non-ideal) whose input voltage  $V_{in}(t) = \cos\omega t$ . Since both the ideal multiplier and the multiplier with feedthrough are linear devices as far as  $V_{in}$  is concerned, (that is, for fixed control voltage) it should be clear that scaling the input signal  $V_{in}(t)$  by a factor of K also scales the output by a factor of K, and that shifting the phase of the input (for example, using  $V_{in}(t) = \cos(\omega t + \phi)$ ) shifts the phase of the output by the same amount. A closely related symbol is  $|V_o|$ , which denotes the peak amplitude of the signal  $V_o(t)$ . With these definitions in mind, we can proceed to develop the properties of the weight model.

An ideal weight (Figure 3) is defined by Equation 1:

$$V_o(t) = I \cos \omega t + Q \sin \omega t \quad (1)$$

where  $V_o(t)$  has been normalized by the assumption that  $V_{in}(t) = \cos \omega t$  as described above.

It is clear that in order to obtain an output of  $A \cos(\omega t + \phi)$ , the I and Q control voltages must be set to

$$I = A \cos \phi \quad Q = -A \sin \phi \quad (2)$$

since

$$\cos \phi \cos \omega t - \sin \phi \sin \omega t = \cos(\omega t + \phi) \quad (3)$$

Incorporating a time delayed feedthrough in parallel with the ideal weight as shown in Figure 3 yields equation 4.

$$V_o(t) = I \cos \omega t + Q \sin \omega t + \gamma \cos(\omega t - \omega \tau_1) \quad (4)$$

We have now unfortunately lost the simple relationship between the output amplitude and phase and the control voltage that is expressed by equation 2, however, we can recover it by making a simple change of variables. We can set the in-phase and quadrature terms to cancel the feedthrough completely at any single frequency. The equation we have to solve is equation 5 in which  $\omega_o$  is the frequency at which the feedthrough is being cancelled.

$$I_o \cos \omega_o t + Q_o \sin \omega_o t = -\gamma \cos(\omega_o t - \omega_o \tau_1) \quad (5)$$

Using the identity of equation 3, we obtain:

$$(\cos \omega_o \tau_1) \cos \omega_o t - (\sin \omega_o \tau_1) \sin \omega_o t = \cos(\omega_o t - \omega_o \tau_1) \quad (6)$$



hence setting  $I_o = -\gamma \cos \omega_o \tau_1$  and  $Q_o = -\gamma \sin \omega_o \tau_1$  will completely null the feedthrough at frequency  $\omega_o$ . If we now make the change of variables  $I = I_o + I_x$ ,  $Q = Q_o + Q_x$  then at frequency  $\omega_o$  we have:

$$V_o(t) = I_x \cos \omega_o t + Q_x \sin \omega_o t \quad (7)$$

To obtain an output of  $A \cos(\omega_o t + \phi)$  at frequency  $\omega_o$ , we need only set  $I_x$  and  $Q_x$  as indicated by equation 2. An expression for  $V_o(t)$  that is valid at all frequencies is:

$$V_o(t) = I_x \cos \omega t + Q_x \sin \omega t + (\cos(\omega t - \omega \tau_1) - \cos(\omega t - \omega_o \tau_1)) \quad (8)$$

Insight into the feedthrough model may be obtained by considering Figure 6. This Figure shows that equation 4 may be pictured as a fixed phasor  $(I-jQ)$  plus a small rotating phasor  $\gamma e^{-j\omega \tau_1}$ . Study of the figure for various values of  $I$  and  $Q$  should make clear why some of the measured data (Figures 7-16) shows curvature in only the phase plot, some only in amplitude, some in both, and some in neither. The assumption made in drawing Fig. 6 is that  $\omega$  varies over a range of frequencies small enough so that  $\gamma e^{-j\omega \tau_1}$  sweeps out less than  $1/2$  radian.

Let us suppose now that  $\omega_o$  is the center of the frequency band of interest. Although it will not be clear until the next section why we are interested in this, let us compute the slope at  $\omega_o$  of both the amplitude and phase of the frequency response of the ideal weight with feedthrough.

The amplitude slope in dB/rad/sec is proportional to:

$$\frac{\partial \ln |V_o|^2}{\partial \omega} = \frac{1}{|V_o|^2} \frac{\partial |V_o|^2}{\partial \omega} \quad (9)$$

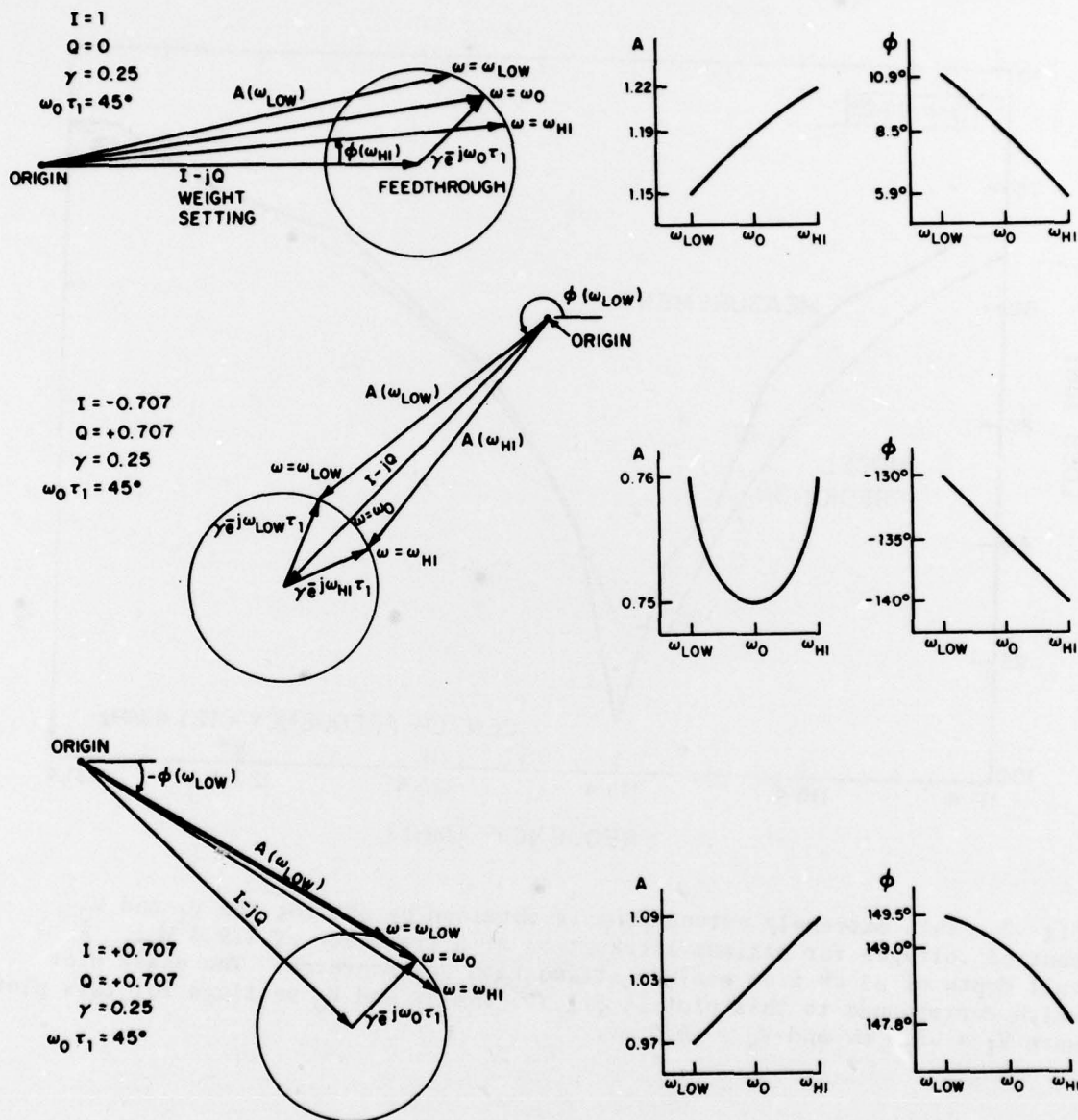


Fig. 6. Graphical illustration of feedthrough.

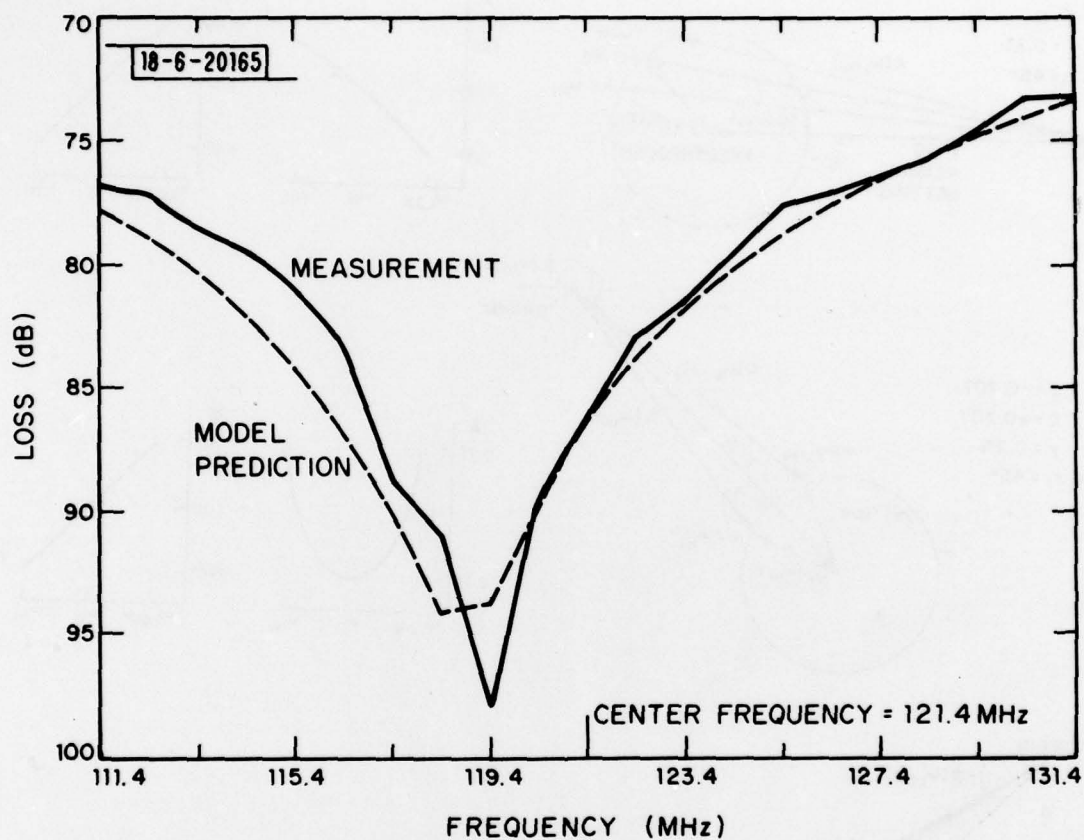


Fig. 7. This extremely curved plot is obtained by setting the  $V_I$  and  $V_Q$  control voltages for maximum attenuation at a frequency of 119.4 MHz. A null depth of 85 dB from minimum attenuation was achieved. The phase plot which corresponds to this plot is Fig. 8. The  $V_I$  and  $V_Q$  settings for this plot were  $V_I = -5.0$  mv and  $V_Q = -6.0$  mv.

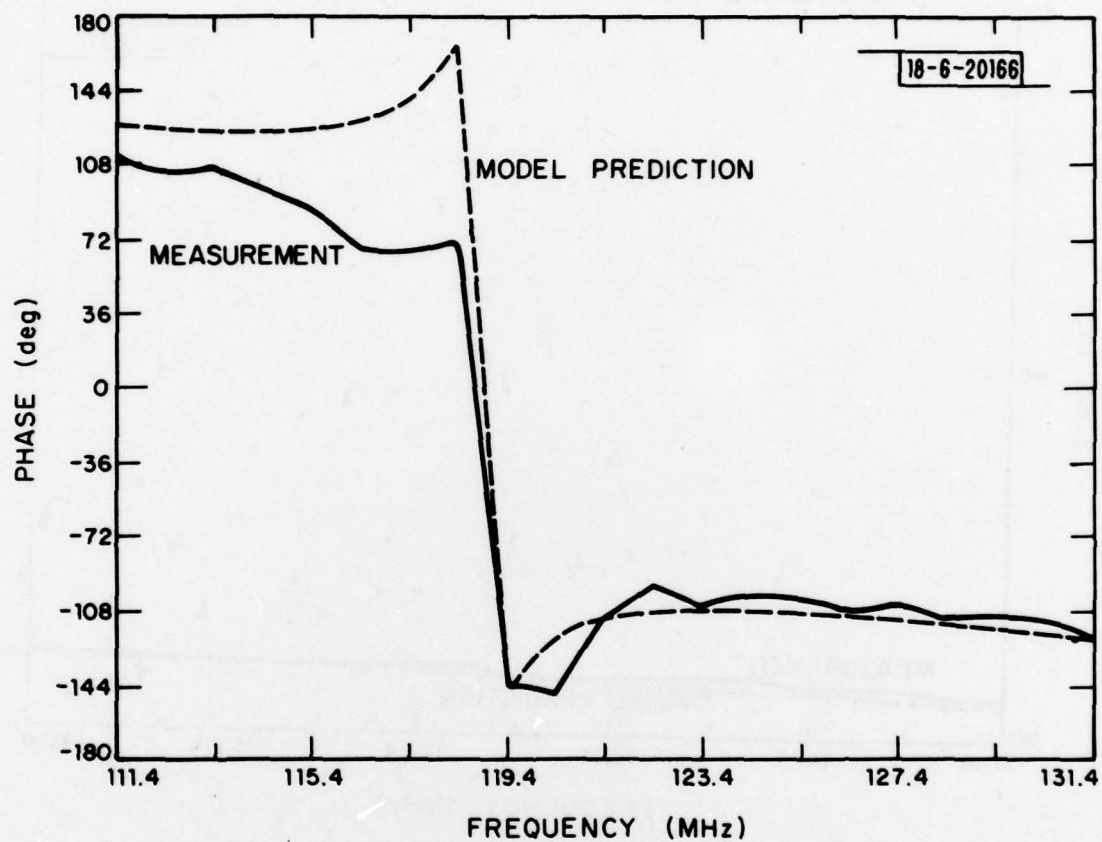


Fig. 8. This is the phase plot which goes with Fig. 7. Note the large phase change in the vicinity of the null.



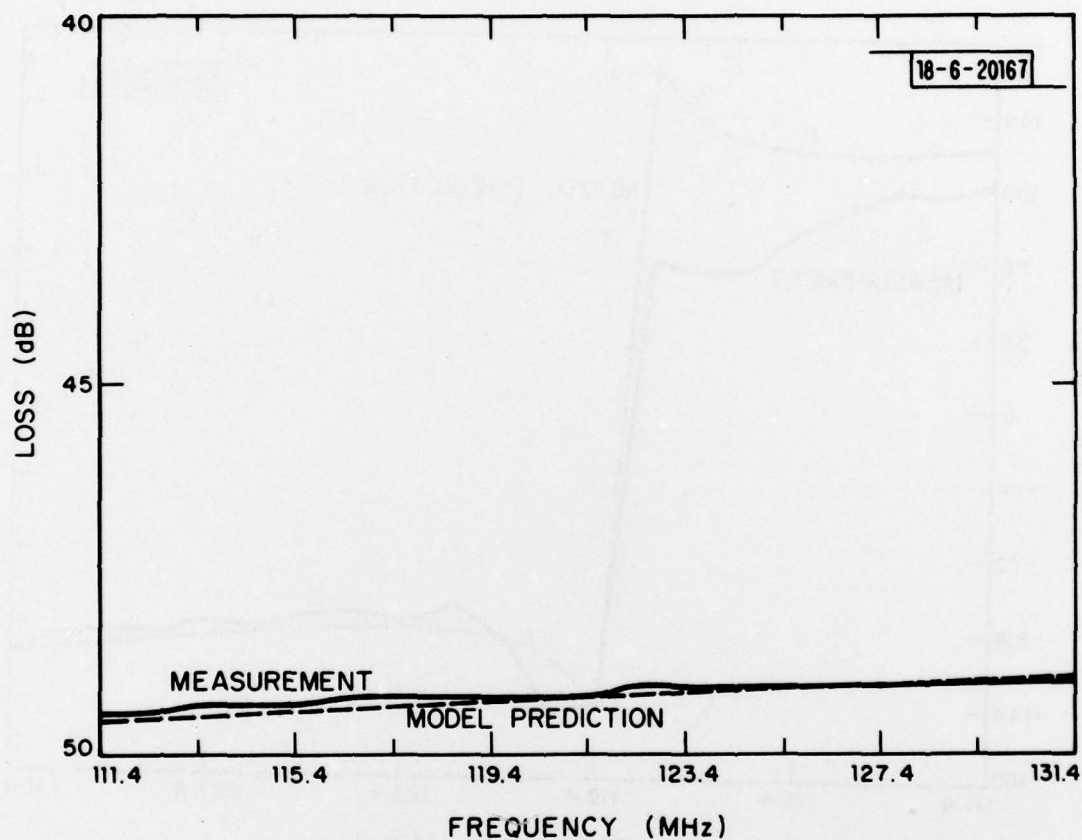


Fig. 9. This almost straight-line plot is obtained by setting  $V_I$  and  $V_Q$  for an attenuation at center frequency at 49.20 dB and a phase at center frequency of  $-47.2^\circ$ . This plot would be suitable for curve fitting since it and its corresponding phase plot (Fig. 10) are so straight. The  $V_I$  and  $V_Q$  settings were  $V_I = 10$  mv and  $V_Q = 0.0$  v.

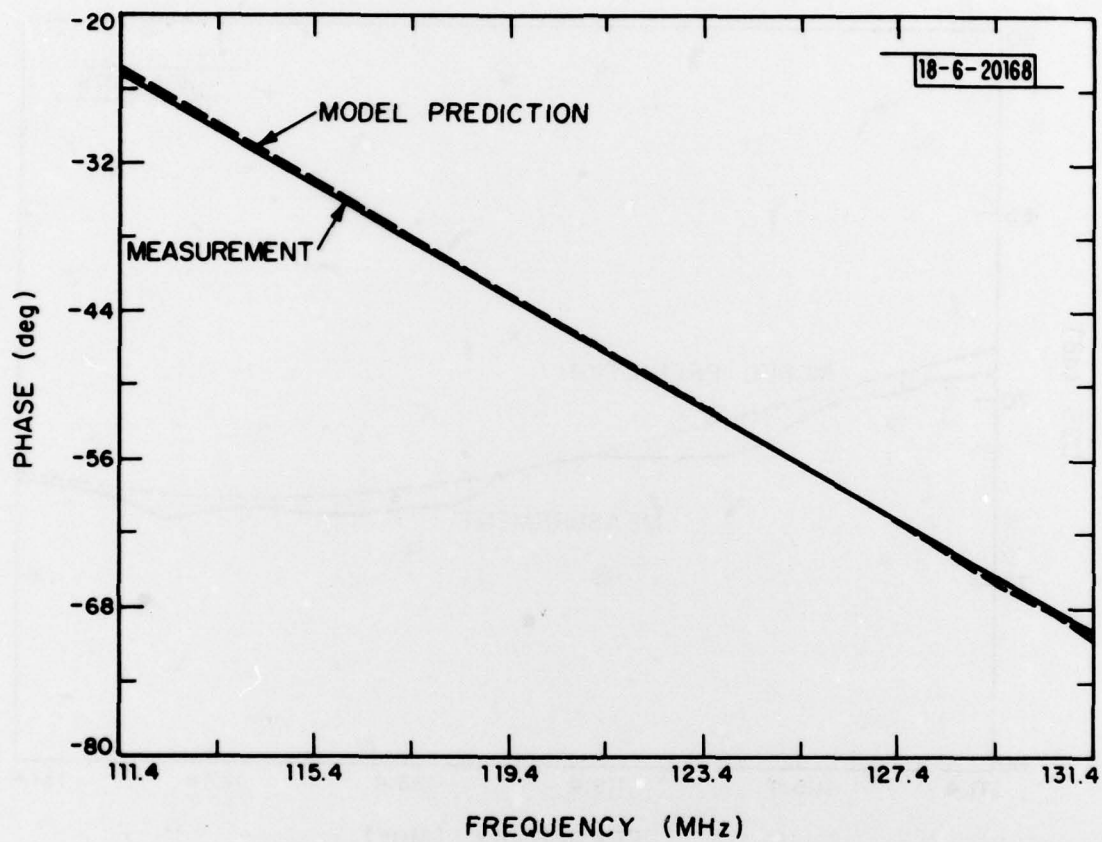


Fig. 10. This phase plot corresponds to the amplitude plot of Fig. 9. Nearly all of the phase slope here is due to the overall delay ( $\tau_2$ ).

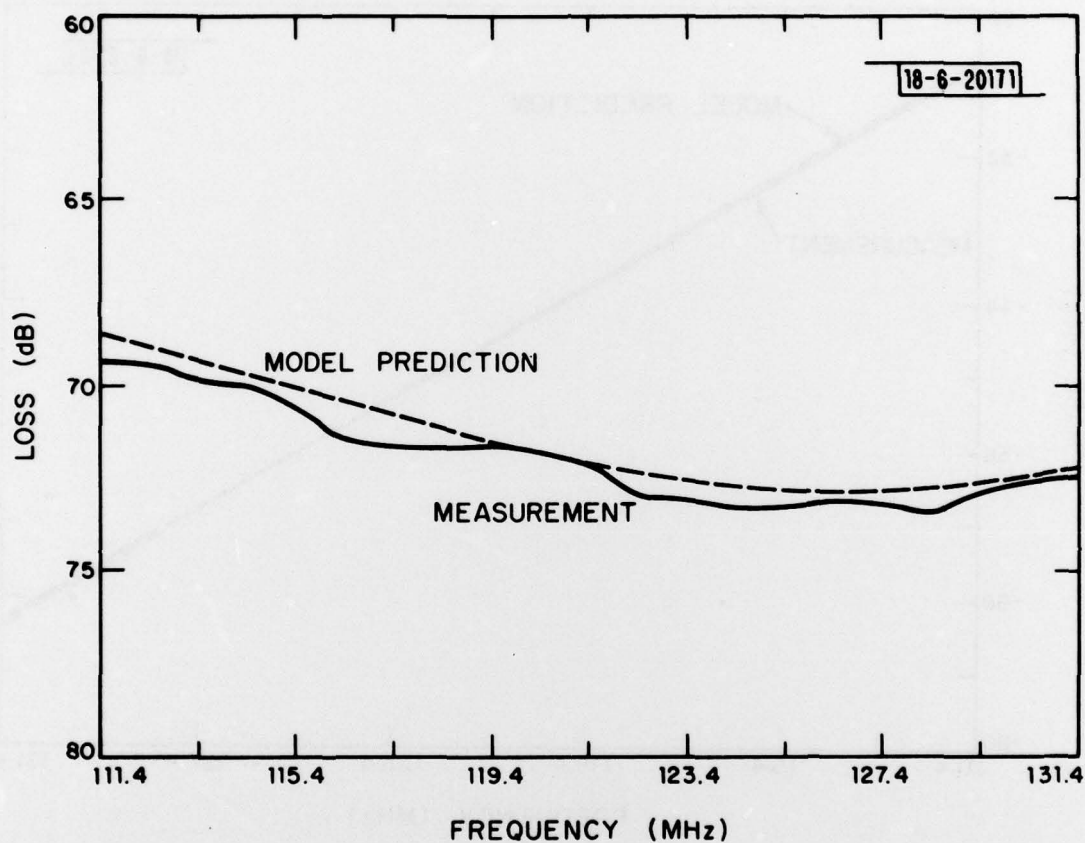


Fig. 11. This amplitude plot exhibits a fair amount of curvature and hence would not be suitable for curve fitting. However, it illustrates the predictive powers of the model nicely as does its companion phase plot (Fig. 12). The  $V_I$  and  $V_Q$  settings here were  $V_I = -6$  mv and  $V_Q = -6$  mv.

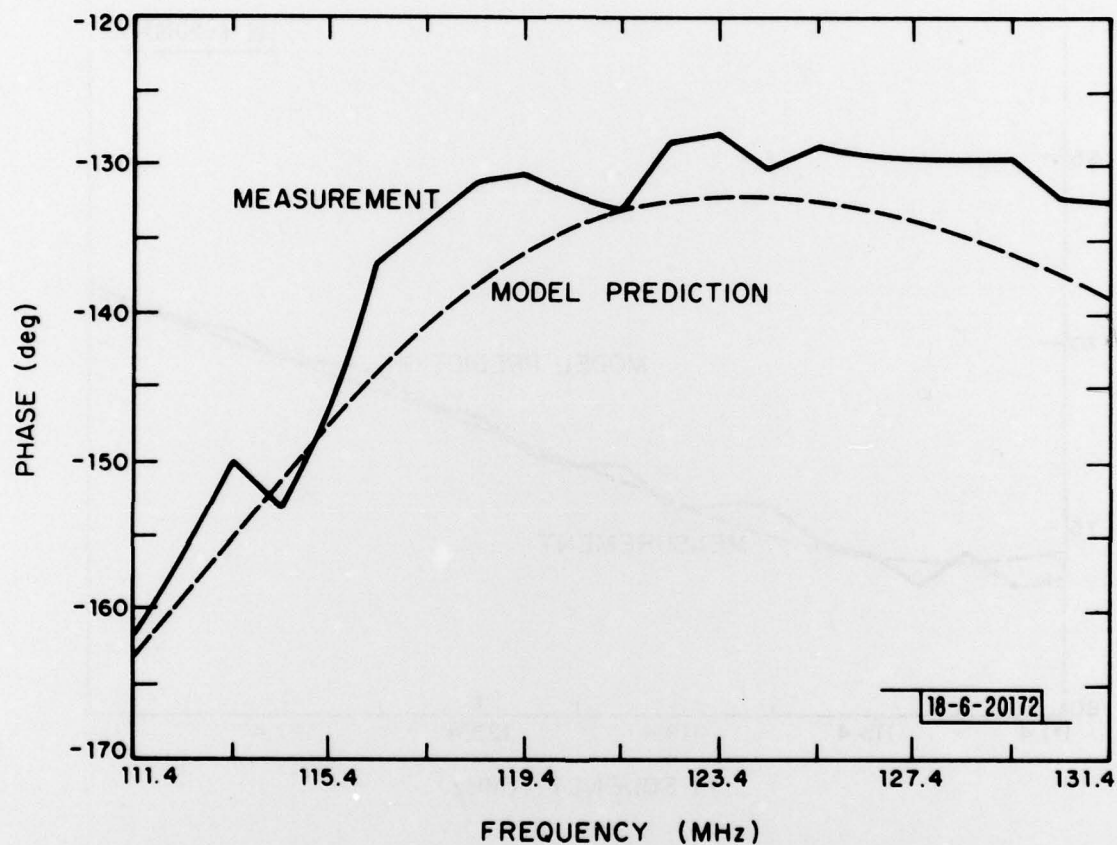


Fig. 12. This phase plot corresponds to the amplitude plot of Fig. 11. Note how noisy the phase measurement is when the attenuation is great.



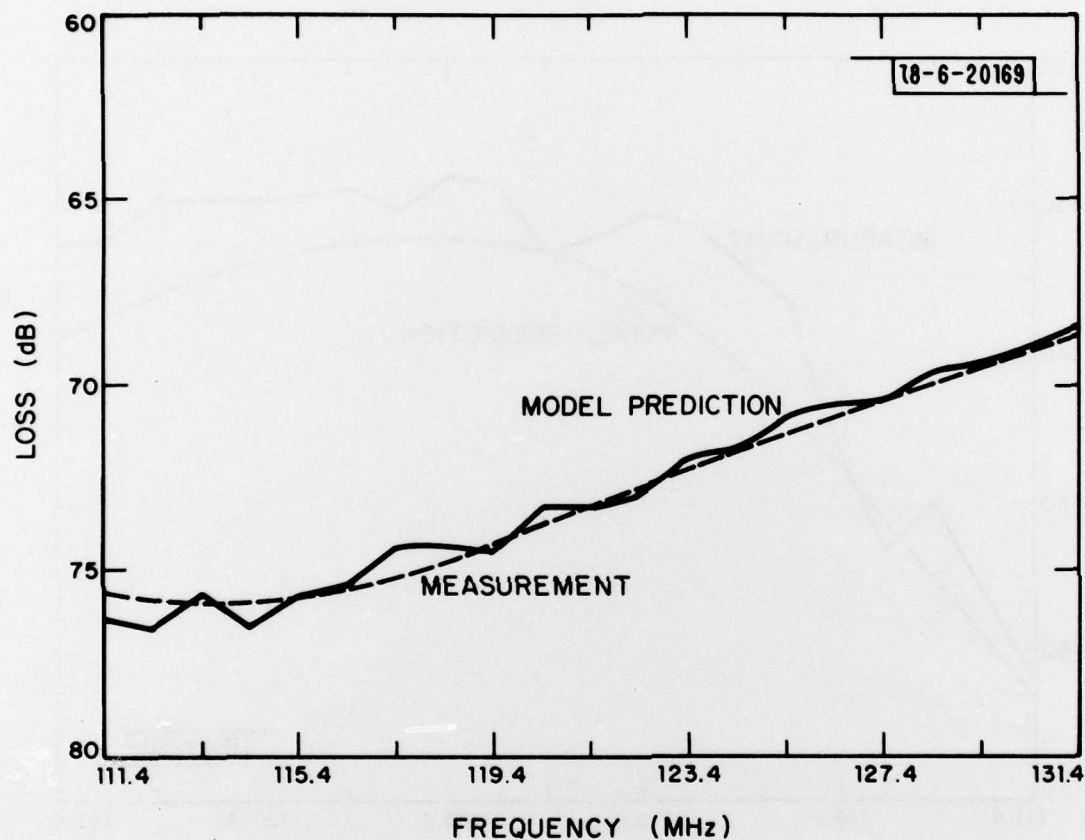


Fig. 13. This amplitude plot is sufficiently straight to be used for curve fitting, however, its corresponding phase plot (Fig. 14) is not straight at all. The control voltages are  $V_I = -5.0$  mv and  $V_Q = -5.0$  mv. Note how much difference a 1 mv change in  $V_I$  and  $V_Q$  can make (compare with Figs. 11 and 12).

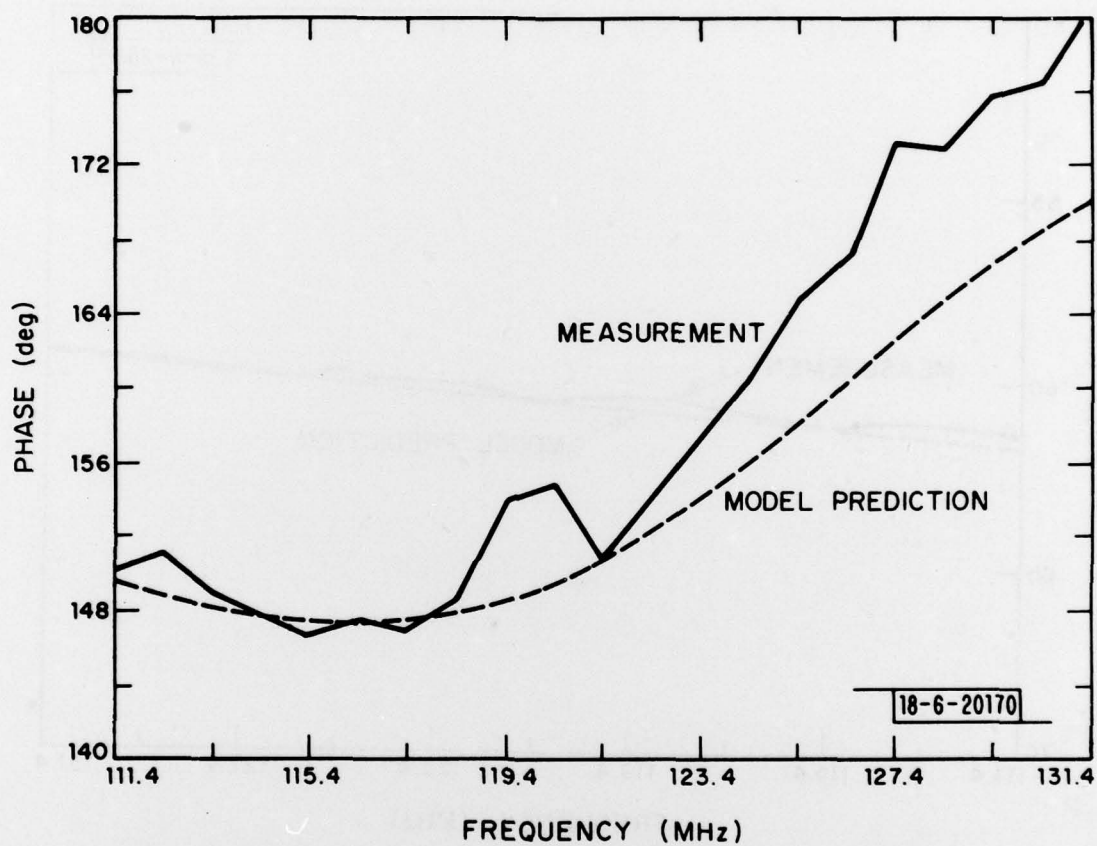


Fig. 14. This phase plot corresponds to the amplitude plot of Fig. 13.

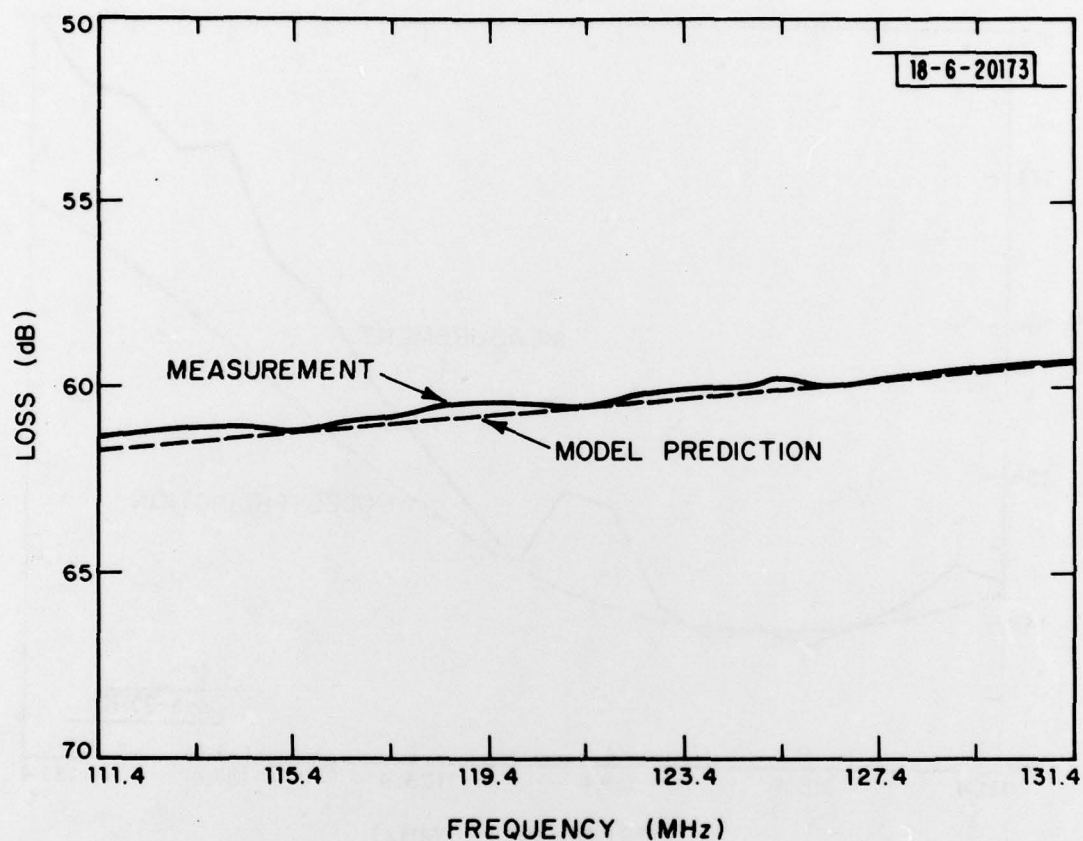


Fig. 15. This amplitude plot and its corresponding phase plot (Fig. 16) are straight enough to be used in the curve fitting procedure. The control voltages were  $V_I = -5.0$  mv and  $V_Q = -1.0$  mv.

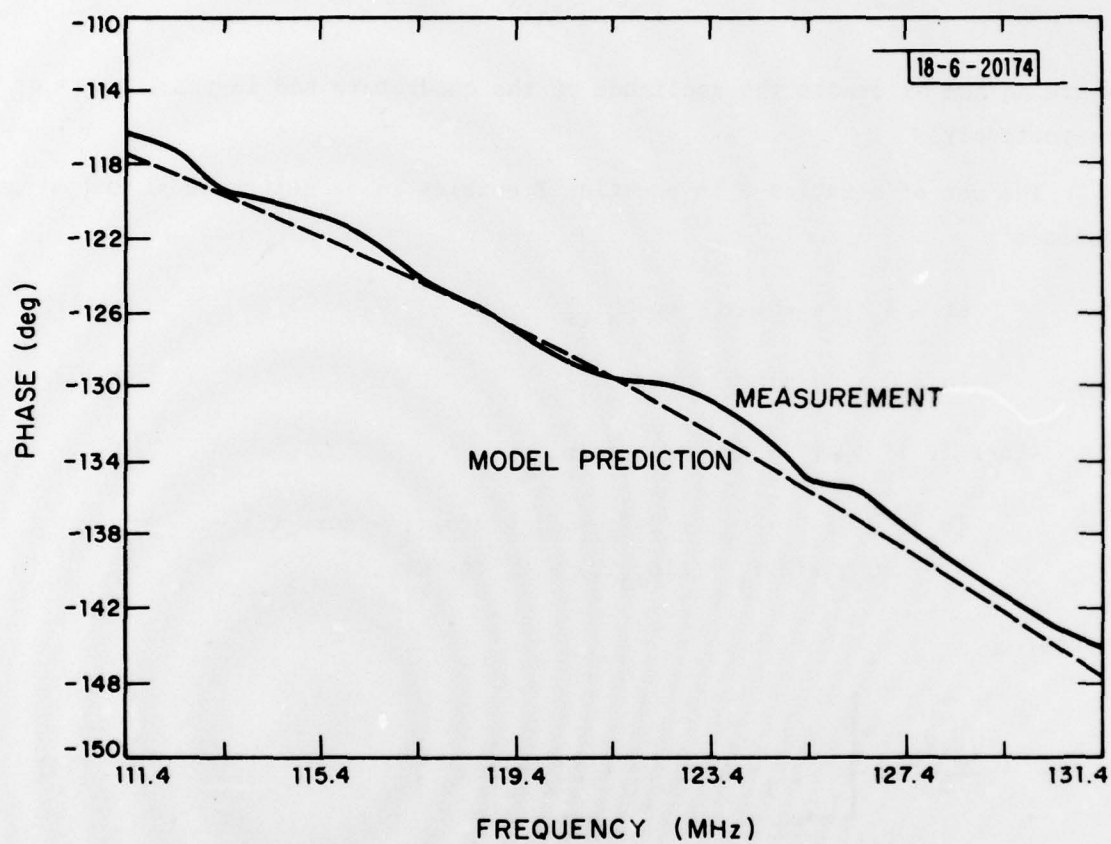


Fig. 16. This phase plot corresponds to the amplitude plot of Fig. 15. Note that this phase measurement is less noisy than those made at greater attenuation (See Figs. 12 and 14).



and the phase slope in deg/rad/sec is proportional to

$$\frac{-\partial \tan^{-1}(Aq/Ai)}{\partial \omega} = \frac{-1}{1 + Aq^2/Ai^2} \frac{\partial (Aq/Ai)}{\partial \omega} \quad (10)$$

where Aq and Ai denote the amplitude of the quadrature and in-phase parts of  $V_o$  respectively.

The use of equation 3 in equation 8 enables us to determine Ai and Aq as follows:

$$Ai = I_x + \gamma(\cos \omega \tau_1 - \cos \omega_o \tau_1) \quad (11)$$

$$Aq = Q_x + \gamma(\sin \omega \tau_1 - \sin \omega_o \tau_1) \quad (12)$$

Since  $|V_o|^2 = Ai^2 + Aq^2$  we obtain:

$$|V_o|^2 = (I_x^2 + Q_x^2 + 2\gamma^2) + 2I_x\gamma(\cos \omega \tau_1 - \cos \omega_o \tau_1) + 2Q_x\gamma(\sin \omega \tau_1 - \sin \omega_o \tau_1) - 2\gamma^2 \cos(\omega \tau_1 - \omega_o \tau_1) \quad (13)$$

Hence:

$$\left. \frac{\partial |V_o|^2}{\partial \omega} \right|_{\omega=\omega_o} = -2I_x\gamma\tau \sin \omega_o \tau_1 + 2Q_x\gamma\tau \cos \omega_o \tau_1 \quad (14)$$

By equation 9 we have:

$$\left. \frac{\partial \ln |V_o|^2}{\partial \omega} \right|_{\omega=\omega_o} = \frac{2\gamma\tau}{I_x^2 + Q_x^2} (Q_x \cos \omega_o \tau_1 - I_x \sin \omega_o \tau_1) \quad (15)$$

Finally, by making the substitution of equation 2 (where A denotes the center frequency amplitude and  $\phi$  the center frequency phase) and using a trigonometric identity we obtain:

$$S_a = \frac{K_a}{2} \frac{\partial \ln |V_o|^2}{\partial \omega} \bigg|_{\omega=\omega_o} = \frac{-K_a \gamma \tau_1}{A} \sin(\omega_o \tau_1 + \phi) \quad (16)$$

where  $S_a$  denotes the amplitude slope and  $K_a$  is a positive constant which is dependent on the units.

By means of equations 10, 11, and 12, we can obtain the phase slope as well. The result is:

$$S_p = \frac{-K_p \partial \tan^{-1}(A_q/A_i)}{\partial \omega} \bigg|_{\omega=\omega_o} = \frac{-K_p \gamma \tau_1}{A} \cos(\omega_o \tau_1 + \phi) \quad (17)$$

It will be seen in the next section that equations 16 and 17 are useful in obtaining numerical values for  $\gamma$  and  $\tau_1$  from swept frequency response data on individual weight circuits.

One issue needs to be briefly discussed. In equation 4 the feedthrough term is shown as being a pure time delay of the input without any additional fixed phase shift. That is, the feedthrough term was shown as  $\gamma \cos(\omega t - \omega \tau_1)$  rather than say  $\gamma \sin(\omega t - \omega \tau_1)$ . It turns out, however, that on the basis of a narrow-band measurement, it is not possible to distinguish between models which differ by only a fixed phase shift in the feedthrough path. For example, if one were to attempt to fit the data to a model with a sinusoidal feedthrough term rather than a cosinusoidal term one would obtain equally as good a fit. The only difference would be different values for  $\gamma$  and  $\tau_1$ ; however, the product  $\gamma \tau_1$  would be unchanged. By narrowband we mean that

$\tau_1(\Delta\omega) < 1/2$ . The measurements of the transconductance weights had a  $\tau_1(\Delta\omega)$  product of approximately  $1/4$ ; therefore, they were narrow band measurements and the above ambiguity applies to them. It is known, however, that only the product  $\gamma\tau_1$  is important in specifying system performance; therefore, the ambiguity is unimportant.

### III. DETERMINATION OF MODEL PARAMETERS

In order to determine the model parameters  $\gamma$ ,  $\tau_1$ , and  $\tau_2$  it is necessary to measure the frequency response of the weight circuits for a variety of weight settings. These measurements can be made with a network analyzer. Figures 7 through 16 are typical plots of the data measured in this way. As done in the previous section, let us denote the center frequency amplitude by  $A$  and the center frequency phase by  $\phi$ .

#### A. Determination of $\tau_2$

From equation 17 we see that if  $A$  is made large enough (compared to  $\gamma\tau_1$ ) then the phase slope due to the feedthrough will be very small. Hence any phase slope which does appear when the weight circuit is set for low attenuation (large  $A$ ) will be due only to the overall time delay,  $\tau_2$ . This phase slope must first be subtracted from all measured phase slopes in order to obtain  $\gamma$  and  $\tau_1$ .

#### B. Solution of Equations 16 and 17

The network analyzer should be used to obtain frequency response data with the control voltages set to obtain several values of  $A$  and  $\phi$ . Values of  $A$  corresponding to attenuations from  $-45$  dB to  $-70$  dB were found to be most useful for the transconductance multiplier weights. These values for  $A$  correspond to the actual attenuation of the device--no correction for average insertion loss is implied. Out of all the measured data some cases will be found to have amplitude and phase characteristics which are approximately straight over the frequency band of interest (see for example Figures 9 and 10). To minimize the effects of noise on slope determination these are the only



cases which should be considered in the solution of Equations 16 and 17. One should take enough measurements to be able to pick about 8 cases with straight amplitude and phase characteristics and a good spread of  $\phi$  values from  $0^\circ$  to  $360^\circ$ . Finally, several measurements should be made with the control voltages set for nearly maximum attenuation. This allows one to verify the correctness of the model by its ability to predict behavior near null (maximum attenuation).

To obtain an initial estimate of  $\gamma$  and  $\tau_1$  we can solve equations 16 and 17 for  $\gamma$  and  $\tau_1$  given values for  $S_a$  and  $S_p$  estimated from measured data for a particular  $A$  and  $\phi$ . (The  $\gamma$  and  $\tau_1$  values determined in this way would be the exact answer if the model were perfect and the measurements noiseless.) This process can be repeated for each measured  $S_a, S_p$  pair. Then the resulting  $\gamma$  and  $\tau_1$  values -- which hopefully will show only a small spread -- may be averaged. These average values can then be used as a starting point in a least squares fitting routine.

Now, let us suppose we have chosen  $S_a, S_p, A, \phi$  from a selected measurement. From Equations 16 and 17 we see that if the model holds then:

$$\frac{S_a}{S_p} = \frac{K_a}{K_p} \frac{\sin(\omega_o \tau_1 + \phi)}{\cos(\omega_o \tau_1 + \phi)} = \frac{K_a}{K_p} \tan(\omega_o \tau_1 + \phi) \quad (18)$$

which indicates that:

$$\tau_1 = \frac{1}{\omega_o} (\tan^{-1} \frac{K_p S_a}{K_a S_p} - \phi) \quad (19)$$

Having obtained  $\tau_1$  we can easily use either Equation 16 or 17 to obtain  $\gamma$ .

Having done this for several cases of  $S_a, S_p, A$ , and  $\phi$ , we may wish to find the  $\gamma$  and  $\tau_1$  values which give the best least-squares fit to all of these several cases. This is the method by which the parameters given in the next section of this report were determined.

### C. Determination of Least Squares Fit

A least squares fit is obtained by picking  $\gamma$  and  $\tau_1$  values which minimize a criterion of goodness. Adequate results were obtained using the criterion:

$$E = \sum_{\substack{8 \text{ selected} \\ \text{measurements}}} \left[ (S_a^p - S_a^m)^2 / (S_a^m)^2 + (S_p^p - S_p^m)^2 / (S_p^m)^2 \right] \quad (20)$$

where  $S_a^p$ ,  $S_p^p$  denote the predictions which the model makes for  $S_a$  and  $S_p$  when given a value of  $\gamma$  and of  $\tau_1$ , and  $S_a^m$ ,  $S_p^m$  denote the measured values. It should be recognized that  $S_a^m$  and  $S_p^m$  are functions of  $A$  and  $\phi$ , that  $S_a^p$  and  $S_p^p$  are functions of  $\gamma$ ,  $\tau_1$ ,  $A$ , and  $\phi$  (see equations 16 and 17), and that the summation in Equation 20 is over a set of  $A$  and  $\phi$  values.

A strategy for picking  $\gamma$  and  $\tau_1$  to minimize  $E$  is to allow  $\gamma$  to vary over a reasonable range in a computer program loop and manually enter successive values of  $\tau_1$  to minimize  $E$ . Each run of the program uses a single  $\tau_1$  value and tries each  $\gamma$  value in the loop to find which one minimizes  $E$ . The program then prints out the  $\gamma$  value which minimized  $E$  for the given  $\tau_1$  value and what value of  $E$  resulted. The user then picks another value of  $\tau_1$  and reruns the program -- stopping when  $E$  is deemed small enough. This strategy avoids the complexity of a gradient search method and was found not to take too much operator time.

### IV. MEASUREMENTS

Extensive measurements of the frequency response of transconductance multiplier weight circuits #2 and #3 were made on a Hewlett-Packard network analyzer. Values of  $\gamma$ ,  $\tau_1$ , and  $\tau_2$  were determined as described above and then plugged into a program which simulated the entire model shown in Figure 5. In both weight circuits very good agreement between model and measurement was obtained (see Figures 7-16).

The values of  $\gamma$ ,  $\tau_1$ , and  $\tau_2$  determined were as follows:

WEIGHT #2	WEIGHT #3
$\gamma = -7.86 \times 10^{-4}$	$\gamma = -6.52 \times 10^{-4}$
$\tau_1 = 4.66$ nanoseconds	$\tau_1 = 4.15$ nanoseconds
$\tau_2 = 5.7$ nanoseconds	$\tau_2 = 5.7$ nanoseconds

Although the gains of the I and Q paths were not precisely equal, they were never more than about 1 dB apart so it is reasonable to speak of a single insertion loss figure of between 12 and 13 dB for both weight circuits #2 and #3. It should be emphasized that in these plots (as well as in the rest of this report) all attenuation values are the actual attenuation that the circuit produces. There has been no correction for average insertion loss, overall time delay, or any other effects.

A number of plots were made of measured and predicted frequency response. See for example Figures 7 through 16. Solid lines are the measured data and dotted lines are the data predicted by the model. Perhaps the most striking feature of these plots is that even though all the curve fitting was done with straight line data (as shown in Figures 9, 10, 15, 16) the model still manages to predict the measured data even when this data is very curved (as shown in Figures 7, 8, 11, 12, 14). The data is for weight circuit #3.

## V. CONCLUSIONS

As inspection of Figures 7 through 16 will indicate, it is possible to obtain a simple model of the transconductance multiplier weight circuit with parameters determined from measured data on individual circuits. This model allows prediction of antenna nulling system performance and thus allows one to intelligently specify the performance requirements on the weight circuits needed to achieve a specified degree of interference cancellation or conversely to infer from the state-of-the-art in multiplier circuits the amount of interference cancellation achievable with antenna nulling systems due to multiplier limitations.



UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

19 REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER 18 ESD-TR-79-161	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) 6 Characterization of Transconductance Multiplier Weight Circuits for an Adaptive Antenna System	5. TYPE OF REPORT & PERIOD COVERED 9 Technical Note	
7. AUTHOR(s) 20 Bruce C. Levens	6. PERFORMING ORG. REPORT NUMBER Technical Note 1979-23	
9. PERFORMING ORGANIZATION NAME AND ADDRESS Lincoln Laboratory, M.I.T. P.O. Box 73 Lexington, MA 02173	8. CONTRACT OR GRANT NUMBER(s) 15 F19628-78-C-0002	
11. CONTROLLING OFFICE NAME AND ADDRESS Air Force Systems Command, USAF Andrews AFB Washington, DC 20331	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS Program Element No. 63431F Project No. 1227	
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office) Electronic Systems Division Hanscom AFB Bedford, MA 01731	12. REPORT DATE 11 22 Jun 1979	
	13. NUMBER OF PAGES 34 12 33	
	15. SECURITY CLASS. (of this report) Unclassified	
15a. DECLASSIFICATION DOWNGRADING SCHEDULE		
16. DISTRIBUTION STATEMENT (of this Report) 14 Approved for public release; distribution unlimited. TN-1979-23		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES None		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) space communications multiplier weight circuits interference cancellation adaptive antenna transconductance antenna nulling		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Extensive testing of the transconductance multiplier weight circuits designed for the antenna nulling project indicates that a simple time delay feedthrough model fits the measured data very well. Properties of this model are derived and applied to the problem of determining the model parameters from measured data. Plots of actual measurements and model predictions are provided to demonstrate goodness of fit.		

DD FORM 1 JAN 73 1473 EDITION OF 1 NOV 65 IS OBSOLETE

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

207 650

JLB