

AD-A069 108

TECHNICAL  
LIBRARY

AD

AD-E400 299

TECHNICAL REPORT ARLCD-TR-78026

ANALYSIS OF A COMPRESSIBLE FLUID SOFT  
RECOIL (CFSR) CONCEPT APPLIED TO  
A 155 MM HOWITZER

BJORN L. HOFGAARD

MARCH 1979



US ARMY ARMAMENT RESEARCH AND DEVELOPMENT COMMAND  
LARGE CALIBER  
WEAPON SYSTEMS LABORATORY  
DOVER, NEW JERSEY

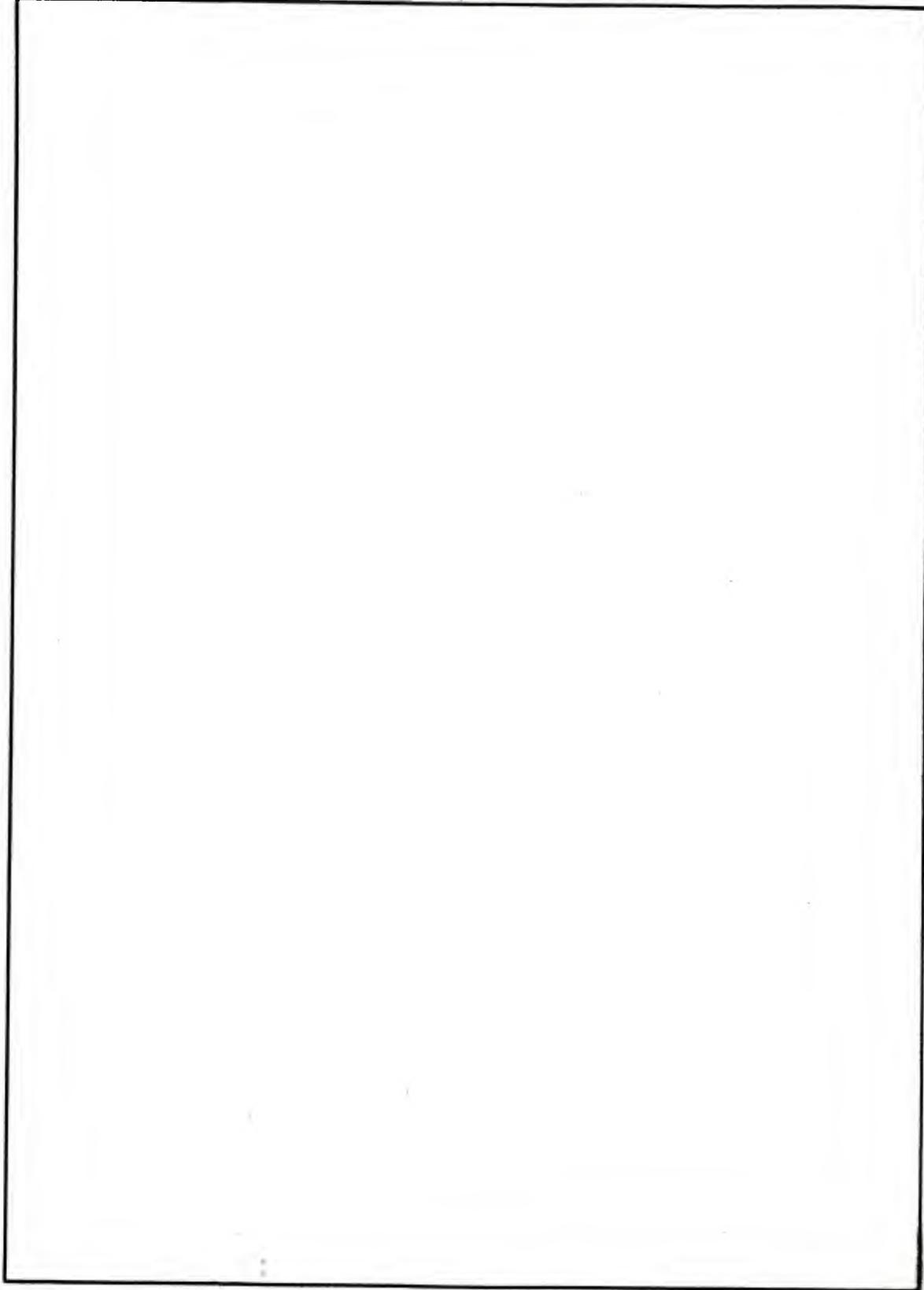
APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED.

## UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER ARLCD-TR-78026	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) ANALYSIS OF A COMPRESSIBLE FLUID SOFT RECOIL (CFSR) CONCEPT APPLIED TO A 155 MM HOWITZER		5. TYPE OF REPORT & PERIOD COVERED
		6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(s) Bjorn L. Hofgaard		8. CONTRACT OR GRANT NUMBER(s) AMCMS Code 662603.H780011
9. PERFORMING ORGANIZATION NAME AND ADDRESS US ARRADCOM Weapons Division, LCWSL Dover, NJ 07801		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS DA Project No. 1W662603AH78
11. CONTROLLING OFFICE NAME AND ADDRESS US ARRADCOM ATTN: DRDAR-TSS Dover, NJ 07801		12. REPORT DATE MARCH 1979
14. MONITORING AGENCY NAME & ADDRESS(if different from Controlling Office) US ARRADCOM Weapons Division, LCWSL Dover, NJ 07801		13. NUMBER OF PAGES 96
16. DISTRIBUTION STATEMENT (of this Report)  Approved for public release, distribution unlimited.		15. SECURITY CLASS. (of this report) UNCLASSIFIED
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Compressible fluid      Liquid flow equations      High pressure seals Bulk modulus      Variable buffer orifice      Cylinder expansion Compressibility      Recoil mechanism      Energy absorbtion Coefficient of flow      Recoil equations		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number)  A mathematical model analyzes a compressible fluid soft recoil (CFSR) concept applied to a 155 mm howitzer. The model addresses the relationships between volume and pressures in a compressible fluid, the forces, deflections, and stresses in the walls of a cylinder which is designed to expand elastically, and the forces and velocities of the recoiling mass. The mathematical model is used with a computer to optimize sizes, pressures, and stresses of the recoil mechanism.		

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)



SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

## **ACKNOWLEDGMENT**

The author wishes to acknowledge the assistance of Mr. Robert H. Coberly in setting up the mathematical model, Mr. Philip D. Benzkofer in running the digital computer program, Mr. Joseph E. Wilson in running the analog computer program, Prof. Arthur P. Boresi for doing the finite elements study, Dr. Rao Yalamanchili for advising on several matters, and Mr. Walter Pape for giving invaluable guidance.

## TABLE OF CONTENTS

	Page No.
Introduction	1
Development and Theory of the Mathematical Model	2
Determination of Cylinder Expansion	2
Effects of Compressibility	8
Recoil Equation	11
Flow Equations	12
Variable Buffer Orifice Area	12
Amount of Energy the System is Able to Store	16
Summary of Mathematical Model	17
Ten Unknown Variables	18
Other Relationships of Important Parametric Design Values	19
Symbols of Basic Engineering Values, Dimensions and Physical Parameters	21
Fixed, Unchangeable Constants	24
Comments Pertaining to the Mathematical Models	25
Results of the Computer Runs	26
Discussion	26
Recoil Cylinder	26
Dow Corning 200 Silicone Fluid	33
Energy Stored	34
Piston	34
Buffers	36
Summary	37
Recommendations	37
References	40
Bibliography	40
Appendixes	
A Breech Force and Breech Energy	41
B Seal Friction	43

C Flow Coefficient	45
D Bulk Modulus	49
E Computer Printouts (Total Program)	51
Distribution List	91

## TABLES

1	Computer run constants	28
2	Calculated data	30
3	Basic engineering values calculated from computer runs; $V_B = 0^*$	31

## FIGURES

1	Compressible fluid soft recoil concept	4
2	Schematic half section of recoil mechanism	5
3	Front buffer	14
4	Rear buffer	15
5	Preliminary CFSR concept study no. 1	27
6	Preliminary CFSR concept study no. 2	39

## INTRODUCTION

Attempts are being made to simplify soft recoil artillery mechanisms, to increase their reliability, and to reduce their need for costly and time-consuming maintenance and rebuild programs.

One approach proposed to accomplish these ends consists of storing the needed operating power or run-up energy partially in an elastically-compressed fluid and partially in elastically-expanding cylinder walls of the recoil mechanism itself. This compressible fluid soft recoil (CFSR) mechanism contains no mechanical springs, gas, replenishers, or recuperators. This results in tremendous simplification. The feasibility of this concept is analyzed in this technical report.

Not much work has been done in this field. Consequently, there are a multitude of unknown parameters and only a limited number of independent physical phenomena taking place which can be set up in mathematical form to solve these unknowns. The unknown parameters include: diameters, lengths, shapes, thicknesses, and orifice sizes; constants, coefficients, physical relationships, and physical properties; pressures, forces, pressure areas, and stresses; fluid flows, volumes, and weights; velocities, required energies, etc. Some of the less critical parameters, therefore, had to be arbitrarily estimated, and some variables had to be fixed.

The basic CFSR concept is thoroughly explained in reference 1. In addition, a brief description is included in the following paragraphs.

In a soft recoil mechanism, a force causes the recoiling mass to move forward before the round is fired. Firing is initiated when a predetermined level of kinetic energy is reached, which is about half of the energy transmitted to the breech by the firing forces. By the time the recoiling mass has returned to the latch position and restored the working energy, all of the firing energy has been dissipated without any large forces being transmitted to the rest of the structure. This permits the size and weight of the remaining structure to be greatly reduced.

In the CFSR concept described by this report, only part of the force which sends the recoiling mass forward before firing is derived from a liquid which has been elastically compressed.

An additional force is imparted by the outer walls of the recoil cylinder which is designed to expand under pressure and store energy elastically.

The expanding cylinder wall theory is a new concept which is evaluated in conjunction with the basic CFSR principle.

## DEVELOPMENT AND THEORY OF THE MATHEMATICAL MODEL

A mathematical model is used primarily to identify and measure significant relationships between controlling parameters. When developing a model, only the most important variables are considered and any secondary effects are generally neglected. Inclusion of too much detail tends to obscure the important relationships, to reduce ability to examine sensitivity to varying input data, and to increase the time required for model development.

In the following paragraphs the most important parameters are optimized (fig. 1, 2).

### Determination of Cylinder Expansion

For a thick-walled cylinder with capped ends, having inner and outer radii of  $r_1$  and  $r_2$ , respectively, and subjected to an internal pressure,  $P_1$ , and an external pressure,  $P_2$ , the tangential, longitudinal, and radial stresses are defined, respectively, by the following equations (ref. 2):

$$\sigma_t = \frac{P_1 r_1^2 - P_2 r_2^2 + (r_1^2 r_2^2 / \rho^2) (P_1 - P_2)}{r_2^2 - r_1^2}, \quad (1)$$

$$\sigma_l = \frac{P_1 r_1^2 - P_2 r_2^2}{r_2^2 - r_1^2}, \quad (2)$$

$$\sigma_r = \frac{P_2 r_2^2 - P_1 r_1^2 + (r_2^2 r_1^2 / \rho^2) (P_1 - P_2)}{r_2^2 - r_1^2}, \quad (3)$$

where  $p$  is the radius to an arbitrary element of the cylinder. The strain in the cylinder is given as

$$\epsilon_t = \frac{1}{E} (\sigma_t - \nu \sigma_r - \nu \sigma_\ell) \quad (4)$$

(stresses are positive in tension) where  $\nu$  is Poissons's ratio (0.287) and  $E$  is the modulus of elasticity  $199.95 \times 10^9$  Pa ( $29 \times 10^6$  psi).

For a thick-walled cylinder with capped ends, having inner and outer radii of  $a$  and  $b$ , respectively, and subjected to an internal pressure only, then;  $P_1 = P$ ,  $r_1 = a$ ,  $r_2 = b$  and  $P_2 = 0$ .

For  $\rho = r_1 = a$

$$\sigma_t = \frac{b^2 + a^2}{b^2 - a^2} P \quad \text{tension,} \quad (5)$$

$$\sigma_\ell = \frac{a^2}{b^2 - a^2} P \quad \text{tension,} \quad (6)$$

$$\sigma_r = P \quad \text{compression,} \quad (7)$$

$$\epsilon = \frac{\Delta a}{a} = \frac{1}{E} \left[ \frac{b^2 + a^2}{b^2 - a^2} P + \nu P - \nu \frac{a^2}{b^2 - a^2} P \right], \quad (8)$$

so

$$\Delta a = P \frac{a}{E} \left[ \frac{b^2 + (1-\nu)a^2}{b^2 - a^2} + \nu \right]. \quad (9)$$

For a thick-walled cylinder with capped ends, having inner and outer radii of  $c$  and  $d$ , respectively, and subjected to an external pressure only, then;  $P_1 = 0$ ,  $P_2 = P$ ,  $r_1 = c$ ,  $r_2 = d$ .

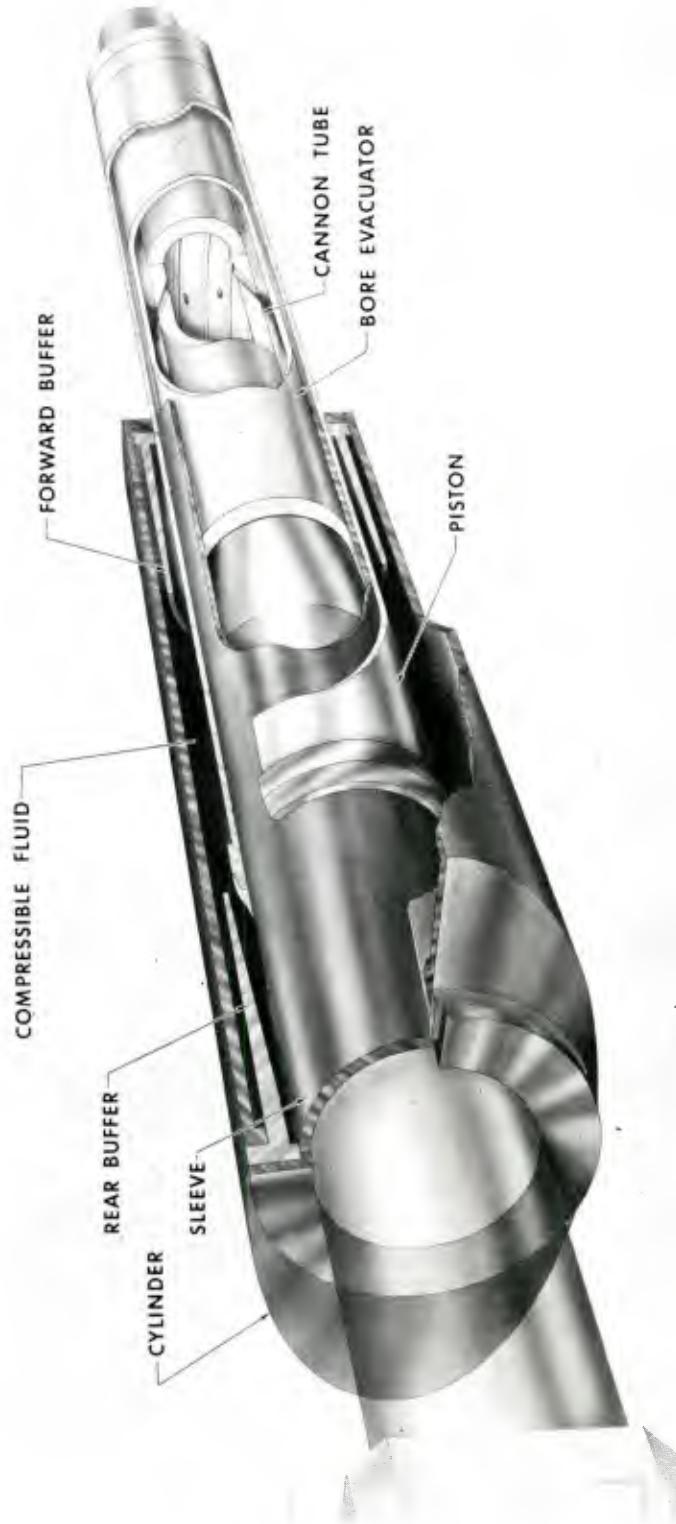


Figure 1. Compressible fluid soft recoil concept.

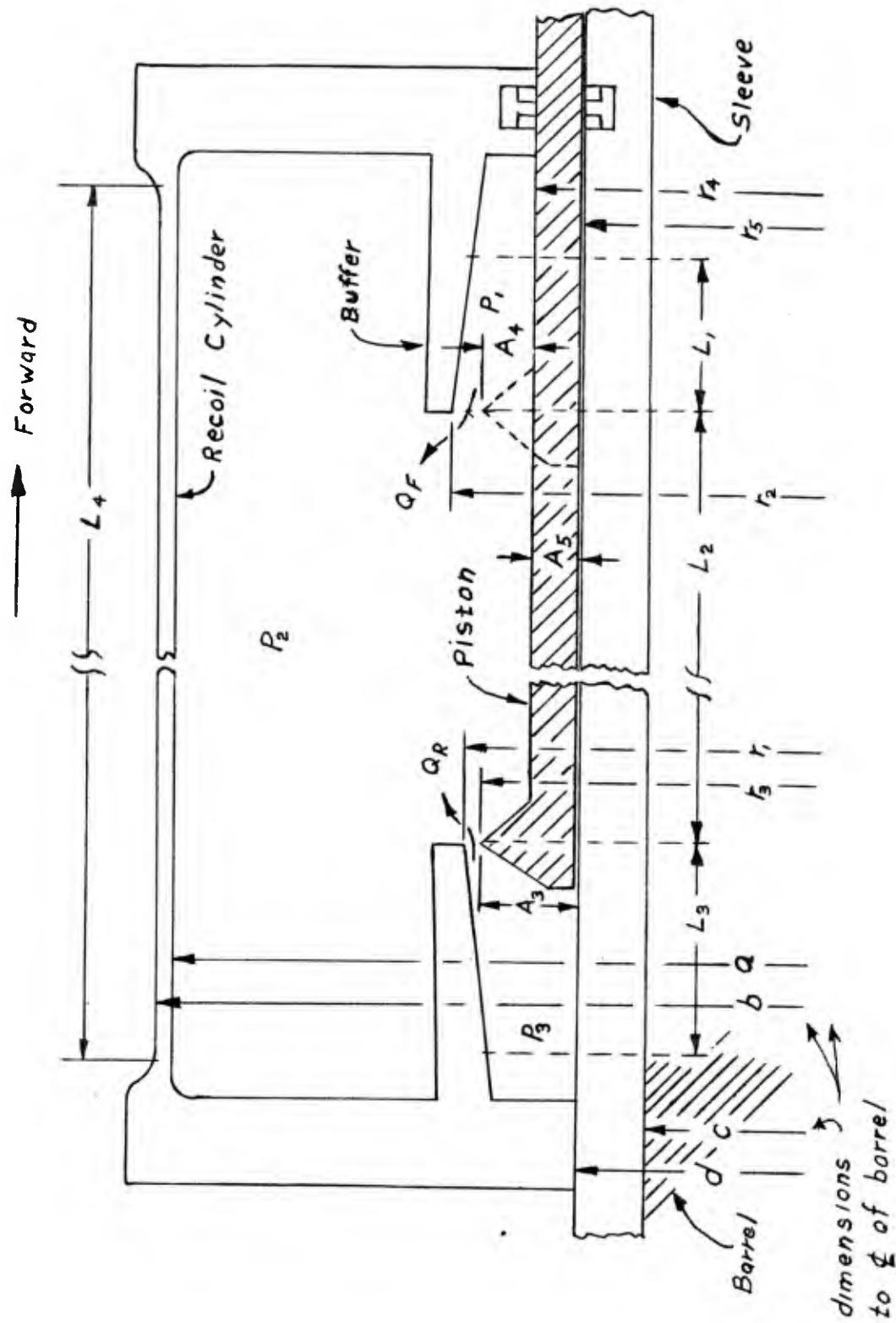


Figure 2. Schematic half section of recoil mechanism.

For  $\rho = r_2 = d$

$$\sigma_t = -P \frac{d^2 + c^2}{d^2 - c^2} \quad \text{tension}, \quad (10)$$

$$\sigma_\ell = -P \frac{d^2}{d^2 - c^2} \quad \text{tension}, \quad (11)$$

$$\sigma_r = P \quad \text{compression}, \quad (12)$$

$$\epsilon_t = \frac{\Delta d}{d} = \frac{-P}{E} \left[ \frac{d^2 + c^2}{d^2 - c^2} - v - v \frac{d^2}{d^2 - c^2} \right], \quad (13)$$

or

$$\Delta d = -P \frac{d}{E} \left[ \frac{d^2(1-v) + c^2}{d^2 - c^2} - v \right]. \quad (14)$$

Define an effective spring rate,  $K_1$  such that

$$2\pi(a+\Delta a) L_1 P_1 = K_1 \Delta a \quad (15)$$

where  $L_1$  is the length of the cylinder on which the internal pressure,  $P_1$  acts.

$$K_1 = 2\pi L_1 P_1 \left( \frac{a}{\Delta a} + 1 \right),$$

but

$$\frac{a}{\Delta a} = \frac{E}{P_1} \frac{1}{\left[ \frac{b^2 + (1-v)a^2}{b^2 - a^2} + v \right]}, \quad (16)$$

thus

$$K_1 = \frac{2\pi L_1 E}{\frac{b^2 + (1-v)a^2}{b^2 - a^2} + v} + 2\pi L_1 P_1. \quad (17)$$

Similarly, define an effective spring rate,  $K_2$  such that

$$2\pi(d+\Delta d) L_2 P_2 = -K_2 \Delta d \quad (18)$$

where  $L_2$  is the length of the cylinder on which the external pressure,  $P_2$  acts.

$$K_2 = 2\pi L_2 P_2 \left( -\frac{d}{\Delta d} - 1 \right), \quad (19)$$

but

$$-\frac{d}{\Delta d} = \frac{E}{P_2} \frac{1}{\left( \frac{d^2(1-v)+c^2}{d^2-c^2} - v \right)} ; \quad (20)$$

thus,

$$K_2 = \frac{2\pi L_2 E}{\frac{d^2(1-v)+c^2}{d^2-c^2} - v} - 2\pi L_2 P_2 . \quad (21)$$

The mass associated with the internal pressure which is being expanded is  $\pi(b^2-a^2)L_1\rho_S$  where  $\rho_S$  is the density of the material.

One equation of motion becomes

$$\pi(b^2-a^2)L_1\rho_S \frac{\ddot{Y}}{2} = \pi L_1 P_1 Y - \left( \frac{2\pi L_1 E}{\left[ \frac{b^2+(1-v)a^2}{b^2-a^2} + v \right]} + 2\pi L_1 P_1 \right) \frac{Y-2a}{2} \quad (22)$$

where  $Y=2(a+\Delta a)$  is the instantaneous value of the inner diameter.  $\Delta a$  and  $\Delta d$  are the radial expansions of the  $a$  and  $d$  radii.

and  $\Delta a = \frac{Y-2a}{2}$ ,  $\ddot{Y} = 2\Delta\ddot{a}$ .

Rewriting the equation of motion:

$$(b^2 - a^2) \rho_S \ddot{a} = P_1 2a + P_1 2\Delta a - \frac{E 2 \Delta a}{\left[ \frac{b^2 + (1-v)a^2}{b^2 - a^2} + v \right]} - P_1 2\Delta a \quad (23)$$

or

$$(b^2 - a^2) \rho_S \ddot{a} = 2aP - \frac{2\Delta a E}{\left[ \frac{b^2 + (1-v)a^2}{b^2 - a^2} + v \right]} . \quad (24)$$

Similarly

$$(d^2 - c^2) \rho_S \ddot{d} = -2dP - \frac{2\Delta d E}{\left[ \frac{d^2(1-v)+c^2}{d^2 - c^2} - v \right]} . \quad (25)$$

### Effects of Compressibility

The physical volume ( $V$ ) available to the fluid at any instant is:

$$V = \frac{\pi}{4} (Y^2 - Z^2) L_4 + A_5 (X - L_1 - L_2 - L_3) + V_0 - V_B . \quad (26)$$

$X$  is the distance of piston travel from its rearmost position in the rear buffer (see Variable Buffer Orifice Area). See Symbols of Basic Engineering Values, Dimensions, and Physical Parameters for definitions of symbols. By definition, the bulk modulus,  $\beta$ , is:

$$\beta = - \frac{\Delta P}{\Delta V/V} \quad (27)$$

or

$$\Delta P = - \frac{\Delta V}{V} \beta . \quad (28)$$

In differential form with respect to time, t;

$$\frac{dP}{dt} = -\frac{\beta}{V} \frac{dV}{dt} \quad (29)$$

but

$$\frac{dV}{dt} = \frac{\pi L_4}{2} (\dot{Y}\dot{Y} - \dot{Z}\dot{Z}) + A_5 \dot{X} \quad (30)$$

Then the basic equation becomes:

$$\dot{P}_2 = -\beta \frac{\frac{\pi L_4}{2} (\dot{Y}\dot{Y} - \dot{Z}\dot{Z}) + A_5 \dot{X}}{\frac{\pi L_4}{4} (Y^2 - Z^2) + A_5 (X - L_1 - L_2 - L_3) + V_0 - V_B} \quad (31)$$

$$Y = 2a + 2\Delta a \quad Z = 2d + 2\Delta d$$

$$\dot{Y} = 2\Delta \ddot{a} \quad \dot{Z} = 2\Delta \ddot{d} \quad * \beta = NP + 132,500$$

$$Y^2 = 4(a + \Delta a)^2 \quad Z^2 = 4(d + \Delta d)^2$$

The desired equation becomes:

$$*\dot{P}_2 = - (NP_2 + 132,500) \frac{2\pi L_4 [(a + \Delta a) \Delta \ddot{a} - (d + \Delta d) \Delta \ddot{d}] + A_5 \dot{X}}{\pi L_4 [(a + \Delta a)^2 - (d + \Delta d)^2] + A_5 (X - L_1 - L_2 - L_3) + V_0 - V_B} \quad (32)$$

\*This formula is expressed in English units.

This equation applies only to  $P_2$  during normal fire, when the piston is between the buffers. If the buffer volumes are added to the equation, this same equation can be applied to  $P_2$  for all three firing conditions (normal, misfire, and cook-off). These additional volumes can be prefixed with controlling functions  $J$ ,  $R$ , and  $U$ , to "switch" these portions on or off as the case may be, depending on the location of the piston.

Therefore, the final equation becomes:

$$*\dot{P}_2 = -(NP_2 + 132,500) \frac{2\pi L_4 [(\dot{a} + \Delta a) \Delta \dot{a} - (\dot{d} + \Delta d) \Delta \dot{d}]}{\pi L_4 [(a + \Delta a)^2 - (d + \Delta d)^2] + A_5 (X - L_1 - L_2 - L_3)} \\ + \frac{+RA_5 \dot{X} - J(A_4 \dot{X} + Q_R) + U(A_4 \dot{X} - Q_F)}{+V_0 - V_B - J(A_3 + A_R)X + U(A_4 + A_F)(X - L_1 - L_2 - L_3)} . \quad (33)$$

The equation for the pressure in the front buffer pocket,  $P_1$ , becomes:

$$*\dot{P}_1 = U(P_1 N + 132,400) \frac{(A_4 \dot{X} - Q_R)}{(A_4 + A_F)(L_1 + L_2 + L_3 - X)} \quad (34)$$

---

\*These equations are expressed in English units.

The equation for the pressure in the rear buffer pocket,  $P_3$ , becomes:

$$*\dot{P}_3 = -J(P_3N + 132,500) \frac{(A_3X + Q_R)}{(A_3 + A_R)X} \quad (35)$$

(see figure 2).

$J = 1$  when  $X < L_3$

$J = 0$  when  $X > L_3$

$R = 1$  when  $X > L_3$

$R = 0$  when  $X < L_3$

$U = 1$  when  $X > L_2 + L_3$

$U = 0$  when  $X < L_2 + L_3$

### Recoil Equation

The recoil equation is

$$*M_R \ddot{X} = S(2.69 P_3(.5) + 2.97 P_1) + P_3 A_3 - P_1 A_4 - HB(t) - W_R \sin \gamma \quad (36)$$

where  $M_R$  and  $W_R$  are the mass and weight, respectively, of the recoiling parts.  $B(t)$  is the breech force (see Appendix A), and  $H$  is a control function that initiates  $B(t)$  when  $\dot{X}=V_e$ .  $P_3 A_3$  and  $P_1 A_4$  are forces on the piston.  $*2.69P_3(.5)+2.97P_1$  is seal friction (see Appendix B), and  $S$  is a control function which changes the direction of the friction force when the piston changes direction.  $\gamma$  is the angle of elevation of the weapon.

---

\*These formulas are expressed in English units.

## Flow Equations

The classical flow equations applied to the buffers, become:

$$Q_F = A_F K_F \sqrt{\frac{2g}{\omega} |(P_1 - P_2)|} \quad \text{sign of } P_1 - P_2 = \text{sign of } Q_F \quad (37)$$

$$Q_R = A_R K_R \sqrt{\frac{2g}{\omega} |(P_3 - P_2)|} \quad \text{sign of } P_3 - P_2 = \text{sign of } Q_R \quad (38)$$

where  $Q$  is the rate of flow in  $\text{in}^3/\text{sec}$ ,  $A$  is the orifice area at any instant,  $K$  is coefficient of discharge,  $\omega$  is the specific weight of the fluid, and  $P$  is the pressure.

For development of this formula and a discussion of coefficient of discharge see Appendix C.

### Variable Buffer Orifice Area

#### Front Buffer Area $A_F$

Assuming a straight linear taper (see fig 3):

$$\frac{r_2 - r_3}{L_1} = \frac{r_X - r_3}{L_1 + L_2 + L_3 - X} \quad (39)$$

$$(r_2 - r_3) \left( 1 + \frac{L_2}{L_1} + \frac{L_3}{L_1} - \frac{X}{L_1} \right) = r_X - r_3 \quad (40)$$

$$r_2 + \frac{r_2 L_2}{L_1} + \frac{r_2 L_3}{L_1} - \frac{r_2 X}{L_1} - r_3 - \frac{r_3 L_2}{L_1} - \frac{r_3 L_3}{L_1} + \frac{r_3 X}{L_1} + r_3 = r_X \quad (41)$$

$$G_F = r_2 - r_3 \quad (42)$$

$$A_1 = \pi (r_2^2 - r_3^2) \quad (43)$$

$$r_X = G_F \frac{L_2}{L_1} + G_F \frac{L_3}{L_1} - G_F \frac{X}{L_1} + r_2 \quad (44)$$

$$r_X = \frac{G_F}{L_1} (L_2 + L_3 - X) + r_2 \quad (45)$$

$$A_F = \pi(r_X^2 - r_3^2) \quad (46)$$

$$A_F = \pi \left[ \frac{G_F}{L_1} (L_2 + L_3 - X) + r_2 \right]^2 - r_3^2 \quad (47)$$

$$A_F = \left[ \frac{G_F^2}{L_1^2} (L_2 + L_3 - X)^2 + \frac{2G_F r_2}{L_1} (L_2 + L_3 - X) + r_2^2 - r_3^2 \right] \quad (48)$$

$$A_F = A_1 + \frac{\pi G_F^2}{L_1^2} (L_2 + L_3 - X)^2 + \frac{2\pi G_F r_2}{L_1} (L_2 + L_3 - X) \quad (49)$$

$$A_F = A_1 + \frac{\pi G_F^2}{L_1^2} (L_2 + L_3)^2 + \frac{2\pi G_F r_2}{L_1} (L_2 + L_3) - \left[ \frac{\pi G_F^2}{L_1^2} (L_2 + L_3) + \frac{2\pi G_F r_2}{L_1} \right] X + \frac{\pi G_F^2}{L_1^2} X^2 \quad (50)$$

### Rear Buffer Area $A_R$

Assuming a straight linear taper (see fig 4):

$$\frac{r_1 - r_3}{L_3} = \frac{r_X - r_3}{X} \quad (51)$$

$$r_X = \frac{r_1 X - r_3 X + L_3 r_3}{L_3} \quad (52)$$

$$G_R = r_1 - r_3 \quad (53)$$

$$A_R = \pi(r_X^2 - r_3^2) \quad (54)$$

$$A_R = \pi \left[ G_R \frac{X}{L_3} + r_3 \right]^2 - r_3^2 \quad (55)$$

$$A_R = \pi \left[ G_R^2 \frac{X^2}{L_3^2} + 2G_R \frac{X}{L_3} r_3 + r_3^2 - r_3^2 \right] \quad (56)$$

$$A_R = \frac{2\pi G_R r_3}{L_3} X + \frac{\pi G_R^2}{L_3^2} X^2 . \quad (57)$$

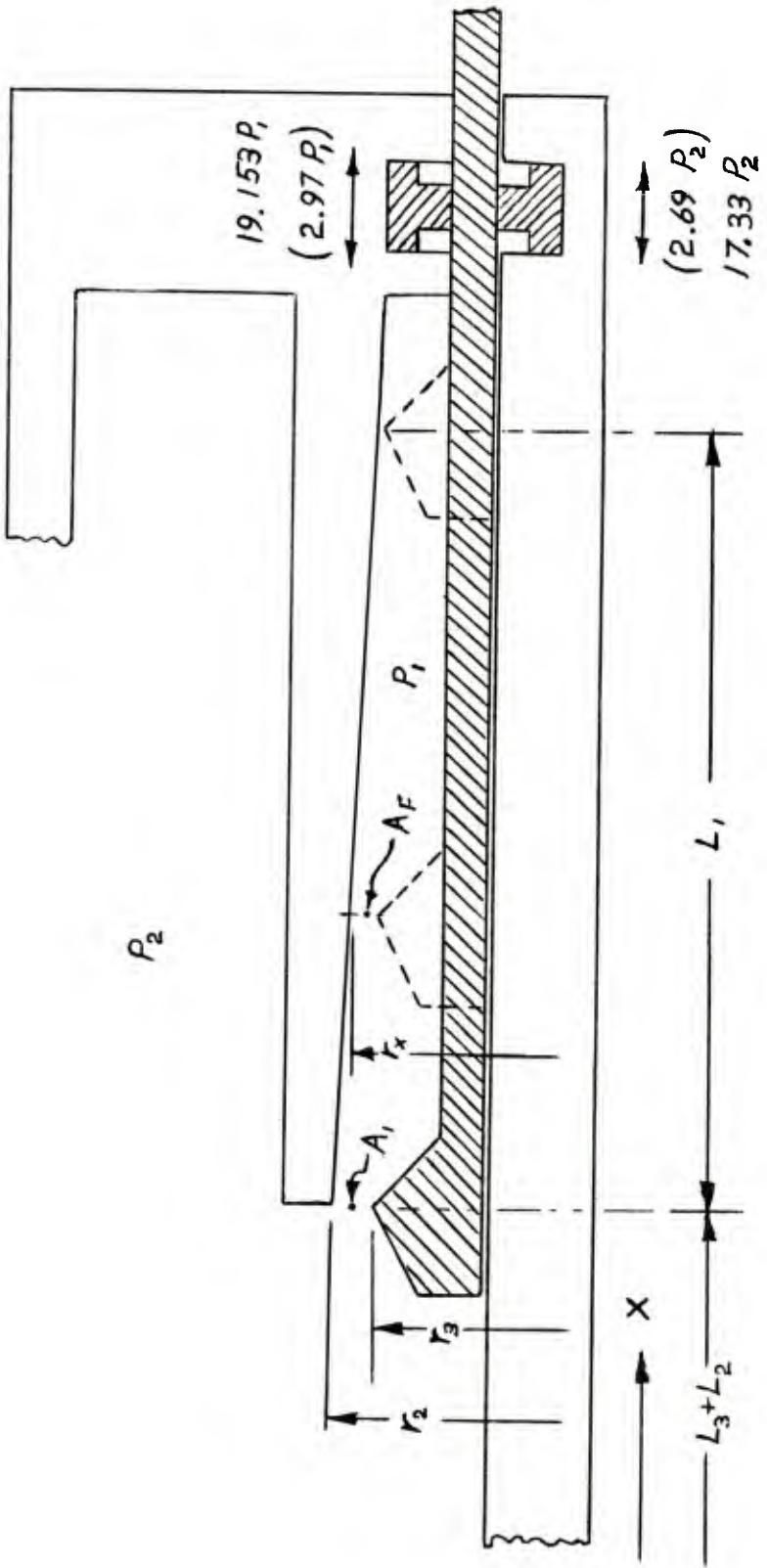


Figure 3. Front buffer.

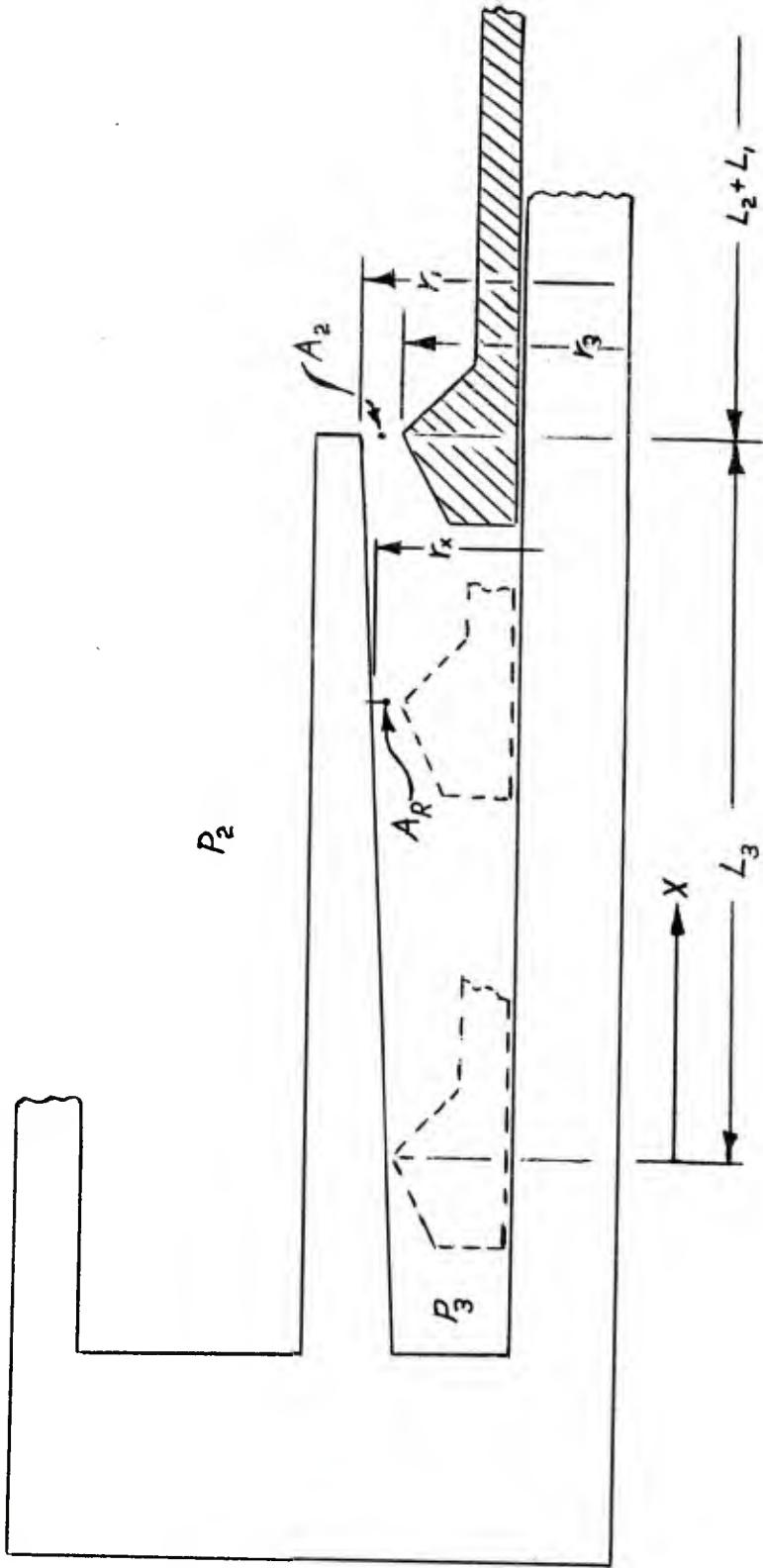


Figure 4. Rear buffer.

### Amount of Energy the System is Able to Store

Energy stored in the compressed fluid,  $E_f$ , is:

$$E_f = \int P dV \quad (58)$$

where  $V$  is the total fluid volume at any instant, and  $P$  is the variable pressure.

$$V = \frac{\pi}{4} (Y^2 - Z^2) L_4 + A_5 (X - L_1 - L_2 - L_3) + V_0 - V_B . \quad (59)$$

$X$  is the distance of piston travel from the rearmost position in the rear buffer.  $A_5$  is the effective net piston area.  $L_4$  is the effective length of both inside ( $Z$ ) and outside ( $Y$ ) cylinders.

$$\Delta P = - \frac{\Delta V}{V} \beta \quad (\text{eqn 28}) \quad (60)$$

$$\frac{dP}{dt} = - \frac{\beta}{V} \frac{dV}{dt} \quad (61)$$

$$dV = - \frac{V}{\beta} \frac{dP}{dt} dt \quad * \beta = NP_2 + 132,500 . \quad (62)$$

Now if these last two equations are substituted into the energy equation one has:

$$*E_f = \int P_2 - \frac{V}{(NP_2 + 132,500)} \frac{dP}{dt} dt \quad (63)$$

or

$$*E_f = - \int \frac{VP_2 \dot{P}_2}{NP_2 + 132,500} dt . \quad (64)$$

Energy stored in the flexible cylinder walls,  $E_c$ , is:

$$E_c = \int 2(a+\Delta a)\pi L_4 P d(\Delta a) + \int 2(d+\Delta d)\pi L_4 P d(\Delta d) , \quad (65)$$

$$Y = 2(a+\Delta a) \quad Z = 2(d+\Delta d)$$

$$\dot{Y} = 2\dot{a} \quad \dot{Z} = 2\dot{d}$$

---

\*These formulas are expressed in English units.

$$E_C = \int Y\pi L_4 P \frac{d(\Delta a)}{dt} dt + \int Z\pi L_4 P \frac{d(\Delta d)}{dt} dt , \quad (66)$$

$$E_C = \pi L_4 \int YP \Delta \dot{a} dt + \pi L_4 \int ZP \Delta \dot{d} dt , \quad (67)$$

and finally,

$$E_C = \frac{\pi L_4}{2} \int Y \dot{Y} P_2 dt + \frac{\pi L_4}{2} \int Z \dot{Z} P_2 dt , \quad (68)$$

or

$$E_C = 2\pi L_4 \int (a + \Delta a) \Delta \dot{a} P_2 dt + 2\pi L_4 \int (d + \Delta d) \Delta \dot{d} P_2 dt . \quad (69)$$

### Summary of Mathematical Model

A summary of the model, using all the physical phenomena taking place in the recoil mechanism, may be expressed by the following ten equations. The equations include the ten unknown variables defined in the next section.

$$Q_F = A_F K \sqrt{\frac{2g}{\omega}} |(P_1 - P_2)| \quad \text{sign of } (P_1 - P_2) = \text{sign of } Q_F \quad (70)$$

$$Q_R = A_R K \sqrt{\frac{2g}{\omega}} |(P_3 - P_2)| \quad \text{sign of } (P_3 - P_2) = \text{sign of } Q_R \quad (71)$$

$$A_F = A_1 + \frac{\pi G_F^2}{L_1^2} (L_2 + L_3)^2 + \frac{2\pi G_F r_2}{L_1} (L_2 + L_3) - \left[ \frac{\pi G_F^2 2}{L_1^2} (L_2 + L_3) + \frac{2\pi G_F r_2}{L_1} \right] X + \frac{\pi G_F^2}{L_1^2} X^2 \quad (72)$$

$$A_R = \frac{2\pi G_R r_3}{L_3} X + \frac{\pi G_R^2}{L_3^2} X^2 \quad (73)$$

$$(b^2 - a^2) \rho_S \Delta \ddot{a} = 2a P_2 - \frac{2E \Delta a}{\left[ \frac{b^2 + (1-v_S)a^2}{b^2 - a^2} + v_S \right]} \quad (74)$$

$$(d^2 - c^2) \rho_S \Delta \ddot{d} = -2d P_2 - \frac{2E \Delta d}{\left[ \frac{d^2(1-v_S) + c^2}{d^2 - c^2} - v_S \right]} \quad (75)$$

$$*\dot{P}_1 = U(P_1 N + 132,500) \frac{(A_4 \dot{X} - Q_R)}{(A_4 + A_F)(L_1 + L_2 + L_3 - X)} \quad (76)$$

$$*\dot{P}_2 = -(N P_2 + 132,500) \frac{2\pi L_4 \left[ (a + \Delta a) \Delta \dot{a} - (d + \Delta d) \Delta \dot{d} \right]}{\pi L_4 \left[ (a + \Delta a)^2 - (d + \Delta d)^2 \right] + A_5 (X - L_1 - L_2 - L_3)} \quad (77)$$

$$+\dot{V}_0 - V_B - J(A_3 + A_R)X + U(A_4 + A_F)(X - L_1 - L_2 - L_3) \quad (77)$$

$$*\dot{P}_3 = -J(P_3 N + 132,500) \frac{(A_3 X + Q_R)}{(A_4 + A_R)X} \quad (78)$$

$$\begin{aligned} J &= 1 \text{ when } X < L_3 \\ J &= 0 \text{ when } X > L_3 \end{aligned}$$

$$\begin{aligned} R &= 1 \text{ when } X > L_3 \\ R &= 0 \text{ when } X < L_3 \end{aligned}$$

$$\begin{aligned} U &= 1 \text{ when } X > L_2 + L_3 \\ U &= 0 \text{ when } X < L_2 + L_3 \end{aligned}$$

$$*M_r \ddot{X} = S(2.69 (.5) P_3 + 2.97 P_1) + P_3 A_3 - P_1 A_4 - H B(t) - W_R \sin \gamma \quad (79)$$

$$\begin{aligned} S &= -1 \text{ when } \dot{X} > 0 \\ S &= +1 \text{ when } \dot{X} < 0 \end{aligned} \quad \begin{aligned} H &= 1 \text{ when } \dot{X} > V_e \\ H &= 0 \text{ when } \dot{X} < V_e \end{aligned}$$

### Ten Unknown Variables

The relationships of the following ten unknown variables are expressed by the mathematical model. Fixing their relationships will unlock other relationships and parameters.

---

\*These equations are expressed in English units.

- $Q_F$  = Fluid flow through annular orifice in front buffer  
 $Q_R$  = Fluid flow through annular orifice in rear buffer  
 $P_1$  = Pressure in front buffer area  
 $P_2$  = Pressure in recoil cylinder  
 $P_3$  = Pressure in rear buffer area  
 $A_F$  = Variable orifice area in front buffer  
 $A_R$  = Variable orifice area in rear buffer  
 $X$  = Distance of piston travel from its innermost position in rear buffer  
 $\Delta a$  = Radial expansion of inside radius of outside recoil cylinder  
 $\Delta d$  = Radial expansion of outside radius of inside sleeve

#### Other Relationships of Important Parametric Design Values

$$A_1 = \pi(r_2^2 - r_3^2) \quad (80)$$

$$A_2 = \pi(r_1^2 - r_3^2) \quad (81)$$

$$A_3 = \pi(r_3^2 - r_5^2) \quad (82)$$

$$A_4 = \pi(r_3^2 - r_4^2) \quad (83)$$

$$A_5 = \pi(r_4^2 - r_5^2) = A_3 - A_4 \quad (84)$$

$$G_F = r_2 - r_3 \quad (85)$$

$$G_R = r_1 - r_3 \quad (86)$$

$$\sigma_t = \frac{(b^2 + a^2)}{(b^2 - a^2)} P_2 \quad (\text{see ref 2}) \quad (87)$$

$$\sigma_a = \frac{a^2}{b^2 - a^2} P_2 \quad (88)$$

$$\sigma_r = -P_2 \quad (89)$$

$$2\sigma_e^2 = (\sigma_t - \sigma_r)^2 + (\sigma_r - \sigma_a)^2 + (\sigma_a - \sigma_t)^2 \quad (90)$$

$$V = \pi(a^2 - d^2)L_4 + A_5[X_0 - (L_1 + L_2 + L_3)] + V_0 - V_B \quad (91)$$

$$V_B = (L_4 - L_2) (R_{av}^2 - r_{ab}^2)\pi \quad (92)$$

$$W_f = V_1 \omega \quad (93)$$

$$*E_f = - \int \frac{VP_2 \dot{P}_2}{NP_2 + 132,500} dt \quad (94)$$

$$E_C = \frac{\pi L_4}{2} \int Y \dot{Y} P_2 dt + \frac{\pi L_4}{2} \int Z \dot{Z} P_2 dt \quad (95)$$

$$*X_0: P = \frac{\pi L_4}{\pi L_4} \left[ \frac{(2a + \Delta a)\Delta a + (2d - \Delta d)\Delta d}{(a^2 - d^2) + A_5[X_0 - (L_1 + L_2 + L_3)]} + A_5(X_0 - X) \right] - \frac{V_B}{V_B + V_0} (NP - 132,500) \quad (96)$$

$$C = \frac{100}{\pi L_4} \left[ \frac{(2a + \Delta a)\Delta a + (2d - \Delta d)\Delta d}{(a^2 - d^2) + A_5[X_0 - (L_1 + L_2 + L_3)]} + A_5(X_0 - X) \right] - \frac{V_B}{V_B + V_0} \quad (97)$$

$$P_{cr} = \frac{\pi^2 EI}{4(L_1 + L_2 + L_3 - X_{min})^2} \quad (\text{Euler's formula}) \quad (98)$$

$$I = \frac{\pi(r_4^4 - r_5^4)}{4} \quad (99)$$

$$P_{pt} = A_5 P_2 \quad (100)$$

or

$$= A_3 P_3 - A_4 P_2 \quad (101)$$

\*These equations are expressed in English units.

## Symbols of Basic Engineering Values, Dimensions and Physical Parameters

- $L_1$  = run-up distance in front buffer
- $L_2$  = run-up distance for normal firing (between buffers)
- $L_3$  = run-up distance in rear buffer
- $L_4$  = length of outside flexible cylinder
- $r_1$  = inside radius of rear buffer at its entrance
- $r_2$  = inside radius of front buffer at its entrance
- $r_3$  = outside radius of piston over orifice cam
- $r_4$  = outside radius of piston sleeve
- $r_5$  = inside radius of piston sleeve
- $a$  = inside radius of outside flexible cylinder
- $b$  = outside radius of outside flexible cylinder
- $c$  = inside radius of sleeve
- $d$  = outside radius of sleeve
- $V_B$  = approximate volume of fluid the buffers displace
- $V_0$  = extra fluid volume in associated container (if necessary)
- $V$  = total fluid volume in recoil cylinder
- $W_f$  = total weight of fluid

$V_e$  = firing velocity of recoil mechanism or maximum run-up velocity

$V_l$  = piston velocity (going into latch)

$V_m$  = minimum piston velocity (misfire only)

$P_1 \text{ max}$  = maximum pressure in front buffer area

$P_2 \text{ max}$  = maximum pressure in recoil cylinder

$P_3 \text{ max}$  = maximum pressure in rear buffer area

$P \text{ min}$  = minimum pressure in any of the volumes

$P \text{ init}$  = initial pressure in all volumes when piston is at a point of latch

$X_0$  = the value of X calculated for the point where all pressures are zero

$X_{P_0}$  = the value of X at the point where  $P_2 = 0$  (taken from computer data)

$X \text{ max}$  = maximum value of X in front buffer

$X \text{ min}$  = minimum value of X in rear buffer

$E_f$  = amount of energy absorbed in the fluid

$E_c$  = amount of energy absorbed in the cylinder walls

$E_{\text{br}}$  = impulse energy at breech

$\text{Imp}$  = maximum firing impulse at breech

$\sigma_t$  = tangential stress in recoil cylinder walls

$\sigma_a$  = axial stress in recoil cylinder walls

$\sigma_r$  = radial stress in recoil cylinder walls

$\sigma_e$  = equivalent stress in recoil cylinder walls

- $A_1$  = annular cross-sectional area between front buffer entrance and piston cam  
 $A_2$  = annular cross-sectional area between rear buffer entrance and piston cam  
 $A_3$  = annular cross-sectional area of piston through cam  
 $A_4$  = difference between areas  $A_3$  &  $A_5$   
 $A_5$  = annular cross-sectional area of piston sleeve through shank  
 $A_{S_1}$  = side pressure area of inside seal  
 $A_{S_2}$  = side pressure area of outside seal  
 $G_F$  = radial gap of front buffer at entrance  
 $G_R$  = radial gap or rear buffer at entrance  
 $P_{cr}$  = critical axial buckling load on end of piston  
 $P_{pt}$  = actual maximum load on end of piston  
 $I$  = moment of inertia of piston in bending  
 $C$  = compression of fluid at latch in %  
 $R_{av}$  = average outside radius of both buffers  
 $r_{av}$  = average inside radius of both buffers

### Fixed, Unchangeable Constants

$g$	= 9.81 m/sec <sup>2</sup> (386.088 in/sec <sup>2</sup> )	acceleration of gravity
$E$	= $199.95 \times 10^9$ Pa ( $29 \times 10^6$ psi)	modulus of elasticity for steel
$\nu_s$	= 0.287	Poisson's ratio for steel
$\rho_s$	= $798.685 \text{ kg sec}^2/\text{m}^4$ (.000,732,994 lb sec <sup>2</sup> /in. <sup>4</sup> )	density of steel
$\omega$	= $939.5 \text{ kg/m}^3$ (.033,942 lb/in. <sup>3</sup> )	specific weight of Dow Corning 200 silicon fluid at atmospheric pressure
$\nu_R$	= .48	Poisson's ratio for Nitrile rubber
$\mu_R$	= .50	dynamic coefficient of friction for rubber
$M_R$	= $300.65 \text{ kg sec}^2/\text{m}$ (16.8355 lb sec <sup>2</sup> /in.)	mass of recoiling parts
$W_R$	= 2948.4 kg (6500 lb)	weight of recoiling parts
$B_{(t)}$	= (see appendix A)	firing impulse at breech (variables)
$N$	= 12.67	tangent of slope of bulk modulus curve
$\beta$	= $NP+9,1356$ ( $NP+132,500$ )	bulk modulus (see appendix D)
$K$	= .95	flow coefficient (see appendix C)
$\gamma$	= $75^\circ$ , $45^\circ$ , $10^\circ$	angle of elevation of weapon
H,J,R. S&U		control functions (see page 18)
${}^*\sigma_{\max}$	= $5.516 \times 10^8$ Pa (80,000 psi)	maximum allowable tensile stress in all cylinder parts (safety factor approximately 2)

\*Assuming steel used is

AISI 4130 or 4140 QQ-S-624 or equal.

Rc 36-40 ultimate tensile stress -  $1.2411 \times 10^8$  Pa (180,000 psi)

Tensile yield stress -  $1.1238 \times 10^8$  Pa (163,000 psi)

## COMMENTS PERTAINING TO THE MATHEMATICAL MODELS

Several mathematical models were developed before the final one shown on the preceding pages was derived. Many computer studies were programmed and run based upon these models. Certain important facts emerged from these studies.

Professor A. P. Boresi of the University of Illinois developed a finite elements model and 26 computer runs were made with this program. The results did not solve all of the problems, since this was only a static model. What was established, however, was how thin the cylinder walls could be without being overstressed. In addition, the magnitude of wall deflection and the approximate amount of energy which the walls would be capable of absorbing were determined.

Since the problem is actually a dynamic problem, another computer program was set up with all the velocities and masses taken into account. Mr. Joe Wilson ran this program on an analog computer. A total of 27 runs were made. Since the analog computer is limited in the quantity of unknowns it can solve for, this study was also restricted in its scope. However, it did establish the run-up velocity and the run-up distance required to achieve that velocity.

The factor which will ultimately determine how well the system works, however, is not the amount of energy stored or the run-up velocity, but if the recoiling mass consistently, but just barely returns to the latch position (with all the breech energy spent). Of course, equally important is the piston force (rod pull) transmitted to the structure. The longer the run-up-distance, the smaller this force may be and the softer the recoil will be. These facts show up clearly in the analog printouts.

Finally, after several attempts, one all-inclusive mathematical model was written which was valid for all three firing conditions (normal, misfire, and cook-off) and for any possible position of the piston. The development of this model is summarized in equations 70 through 79. With this model the overwhelming amount of input and output data had to be limited so as not to obscure the most important relationships. However, a tool was now available which could be used for a trial-and-error type of optimization of the most important parameters. About 100 computer runs were made. Of these, eight were good runs which are summarized in the following paragraph. (Mr. Philip Benzkofer programmed and ran this study on a digital computer.)

## RESULTS OF THE COMPUTER RUNS

Table 1 shows the data which were held constant during the 8 "good" computer runs. One of these runs is shown as appendix E.

Figure 5 is a to-scale drawing based upon the dimensions given in table 1. The drawing is merely a concept study in that no thought was given as to how the mechanism could be broken up into smaller parts for ease of machining and assembly.

Table 2 lists relevant engineering information calculated from some of the computer data.

Table 3 has pressures, stresses, and other optimized values from the computer runs.

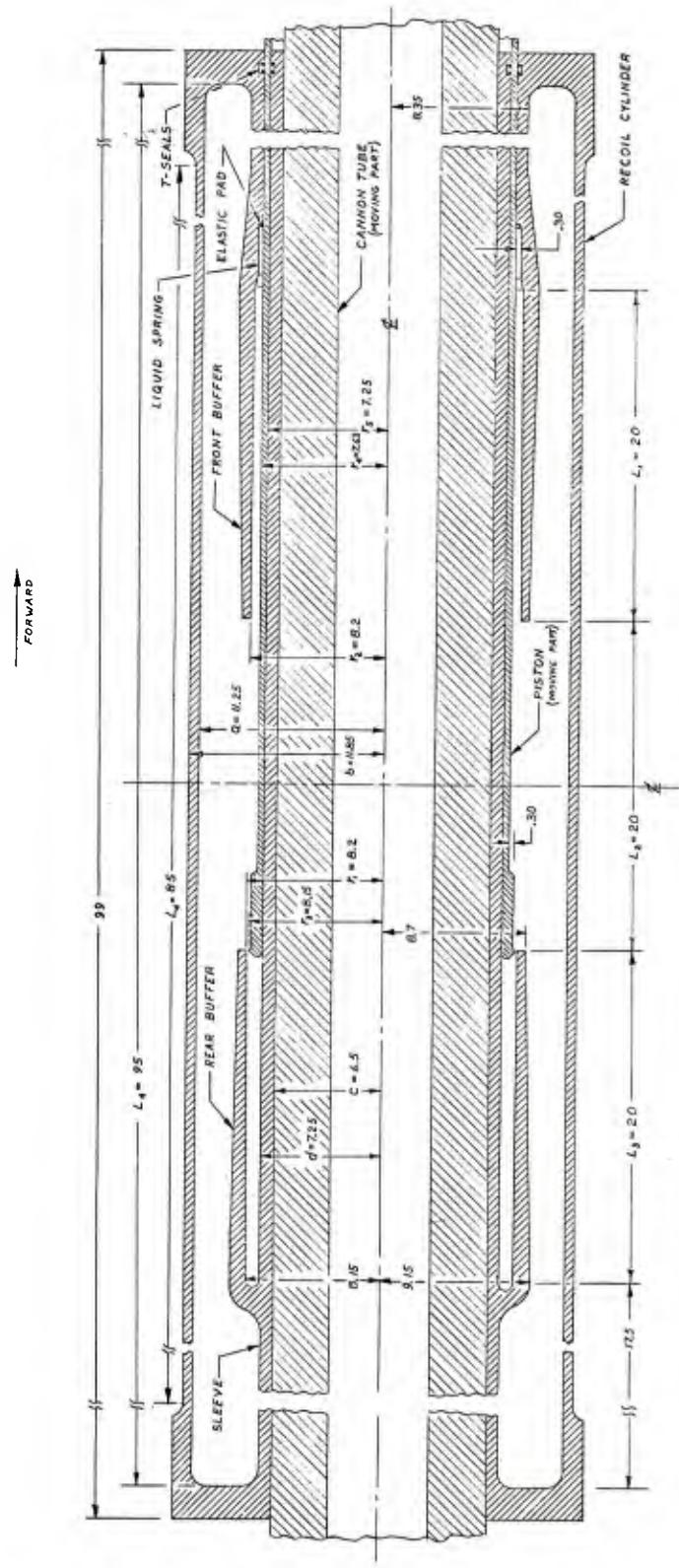
## DISCUSSION

The thought of a compressible liquid recoil system opens up the possibility of a greatly improved and simplified recoil mechanism. There would appear to be a likelihood of storing a large amount of energy in a small space, thus achieving a greatly simplified and "clean" one-principle, one-fluid design. There would be fewer tanks and fewer sub-mechanisms, fewer parts overall. Therefore, the main criterion that has been adhered to throughout the program is simplification.

The RESULTS give a set of ideal dimensions and parameters for the CFSR mechanism which are discussed in detail in the following paragraph.

### Recoil Cylinder (Fig 5)

The outside diameter of the recoil cylinder is 60.20 cm (23.7 in.). The cylinder surrounds (is coaxial with) the 33.02 cm (13-inch) diameter barrel (cannon tube). The effective length of the cylinder is 215.9 cm (85 in.). There remains a rather small space for the fluid. This small cylinder size is made possible only by allowing different parameters for the 3 different firing conditions, i.e., the cylinder still does not hold enough fluid to function properly during misfire. Recognizing how rarely a misfire occurs, this gives the designer a choice of whether or not to design for misfire. The only thing that probably would occur with the system underdesigned for misfire is that the piston would draw a vacuum which would cause cavitation and foaming. This may not be serious if it does not occur too frequently.



**Figure 5. Preliminary CFSR concept study no. 1**  
(Dimensions are given in inches)

**Table 1 Computer run constants**

Parameter	Value mm (in.)
Run-up distance in front buffer ( $L_1$ )	508 mm (20 in.)
Run-up distance for normal firing (between buffers) ( $L_2$ )	508 mm (20 in.)
Run-up distance in rear buffer ( $L_3$ )	508 mm (20 in.)
Length of outside flexible cylinder ( $L_4$ )	2159 mm (85 in.)
Inside radius of rear buffer, at its entrance ( $r_1$ )	208.3 mm (8.20 in.)
Inside radius of front buffer at its entrance ( $r_2$ )	208.3 mm (8.20 in.)
Outside radius of piston over orifice cam ( $r_3$ )	207 mm (8.15 in.)
Outside radius of piston sleeve ( $r_4$ )	193.7 mm (7.625 in.)
Inside radius of piston sleeve ( $r_5$ )	184.2 mm (7.25 in.)
Inside radius of outside flexible cylinder (a)	285.8 mm (11.25 in.)
Outside radius of outside flexible cylinder (b)	301 mm (11.85 in.)
Inside radius of sleeve (c)	165.1 mm (6.50 in.)
Outside radius of sleeve (d)	184.2 mm (7.25 in.)
Outer wall thickness (b-a)	15.2 mm (.60 in.)
Inner wall thickness (d-c)	19.1 mm (.75 in.)
Front buffer gap ( $G_F$ )	1.3 mm (.05 in.)
Rear buffer gap ( $G_R$ )	1.3 mm (.05 in.)
Annular effective front buffer gap area ( $A_1$ )	65.2 mm <sup>2</sup> (2.568 in. <sup>2</sup> )
Annular effective rear buffer gap area ( $A_2$ )	65.2 mm <sup>2</sup> (2.568 in. <sup>2</sup> )

Table 1 (Cont'd)

Parameter	Value mm (in.)
Piston sleeve thickness ( $r_4 - r_5$ )	9.5 mm (0.375 in.)
Annular effective cross sectional piston area ( $A_5$ )	445.1 mm <sup>2</sup> (17.524 in. <sup>2</sup> )
Moment of inertia for cross section of piston (I)	20,187 cm <sup>4</sup> (485 in. <sup>4</sup> )
Maximum critical buckling load on end of piston ( $P_0$ )	5,203,792 kg (11,472,396 lb)
Initial pressure at latch ( $P_{init}$ )	20,684.2 kPa (3000 psi)
Average outside radius of both buffers ( $R_{av}$ )	232.4 mm (9.15 in.)
Average inside radius of both buffers ( $r_{av}$ )	207 mm (8.15 in.)
Approximate volume of buffers, filling oil space ( $V_B$ )	34,113 cm <sup>3</sup> (2,081.7 in. <sup>3</sup> )
Impulse energy at breech ( $E_{br}$ )	597,496 Nn (5,288,287 in.-lb)

Table 2. Calculated data

Extra fluid volume in connected container ( $V_0$ )	0	.164 m <sup>3</sup> (10,000 in. <sup>3</sup> )/ 164 litre (43.29 gal)	0	.164 m <sup>3</sup> (10,000 in. <sup>3</sup> )/ 164 litre (43.29 gal)
Approximate volume of the two buffers ( $V_B$ )	0	0	.0339 m <sup>3</sup> (2,081 in. <sup>3</sup> )	.0339 m <sup>3</sup> (2,081 in. <sup>3</sup> )
Total fluid volume in recoil cylinder (V)	3186 m <sup>3</sup> (19,543 in. <sup>3</sup> )/ 318.6 litre (84.60 gal)	.485 m <sup>3</sup> (29,724 in. <sup>3</sup> )/ 485 litre (128.67 gal)	.2599 m <sup>3</sup> (15,948 in. <sup>3</sup> )/ 259.9 litre (69.04 gal)	.4259 m <sup>3</sup> (26,126 in. <sup>3</sup> )/ 425.9 litre (113.10 gal)
Total weight of fluid ( $W_f$ )	300.9 kg (663.3 lb)	457.6 kg (1,008.9 lb)	245.5 kg (541.3 lb)	402.2 kg (886.8 lb)
*The value of X calculated for the point where all pressures are zero ( $X_0$ )	1209 mm (47.6 in.)	1470.7 mm (57.9 in.)	1117.6 mm (44.0 in.)	1376.7 mm (54.2 in.)
Compression of fluid at latch in % (C)	1.7572 %	1.7624 %	1.7579 %	1.7570 %
C at smallest X	2.6464 %	2.3471 %	2.8476 %	2.4222 %

\*X = distance of piston travel from its innermost position in rear buffer.

Table 1. Basic engineering values calculated from computer runs;  $V_B = 0^*$

Run no.	Mode of fire	1		2		3		4		5		6		7		8	
		$\gamma^*$	$V_0$	0	0	0	0	Normal fire	H = 0 when $\dot{X} \leq V_e$	45	75	0	0	H = 0 Misfire	0	Cook-off	H = 1
$P_1$ max	Pa (psi)	2.863x10 <sup>7</sup>	(4.152)	2.881x10 <sup>7</sup>	(4.179)	2.847x10 <sup>7</sup>	(4.129)	2.936x10 <sup>7</sup>	(4.258)	2.959x10 <sup>7</sup>	(4.281)	2.068x10 <sup>7</sup>	(2.999)	2.418x10 <sup>7</sup>	(3.507)	3.147x10 <sup>7</sup>	(4.564)
$P_4$ max	Pa (psi)	2.863x10 <sup>7</sup>	(4.152)	2.881x10 <sup>7</sup>	(4.179)	2.847x10 <sup>7</sup>	(4.129)	2.936x10 <sup>7</sup>	(4.258)	2.959x10 <sup>7</sup>	(4.281)	2.068x10 <sup>7</sup>	(2.999)	2.408x10 <sup>7</sup>	(3.000)	3.147x10 <sup>7</sup>	(4.564)
$P_a$ max	Pa (psi)	3.833x10 <sup>7</sup>	(5.560)	4.024x10 <sup>7</sup>	(5.837)	3.840x10 <sup>7</sup>	(5.589)	4.273x10 <sup>7</sup>	(6.197)	4.444x10 <sup>7</sup>	(6.445)	2.068x10 <sup>7</sup>	(2.999)	2.068x10 <sup>7</sup>	(3.000)	6.930x10 <sup>7</sup>	(10.060)
$P_a$ min	Pa (psi)	-1.248x10 <sup>8</sup>	(-181)	9.255x10 <sup>6</sup>	(1.342)	8.714x10 <sup>8</sup>	(974)	1.303x10 <sup>8</sup>	(189)	2.241x10 <sup>8</sup>	(325)	-5.792x10 <sup>8</sup>	(-840)	3.034x10 <sup>8</sup>	(44)	1.682x10 <sup>8</sup>	(2.439)
$Q^*$	Pa (psi)	5.514x10 <sup>6</sup>	(79.977)	5.550x10 <sup>6</sup>	(79.977)	5.485x10 <sup>6</sup>	(79.546)	5.650x10 <sup>6</sup>	(82.027)	5.698x10 <sup>6</sup>	(82.654)	3.984x10 <sup>8</sup>	(57.781)	3.984x10 <sup>8</sup>	(57.781)	6.061x10 <sup>8</sup>	(87.912)
$Q^t$	Pa (psi)	2.614x10 <sup>6</sup>	(37.912)	2.631x10 <sup>6</sup>	(38.158)	2.600x10 <sup>6</sup>	(37.708)	2.683x10 <sup>6</sup>	(38.844)	2.702x10 <sup>6</sup>	(39.182)	1.888x10 <sup>8</sup>	(27.389)	1.888x10 <sup>8</sup>	(27.389)	2.875x10 <sup>8</sup>	(41.674)
$Q^a$	Pa (psi)	-2.863x10 <sup>7</sup>	(-4.152)	2.881x10 <sup>7</sup>	(4.179)	-2.848x10 <sup>7</sup>	(-4.130)	-2.938x10 <sup>7</sup>	(-4.258)	-2.959x10 <sup>7</sup>	(-4.291)	-2.068x10 <sup>7</sup>	(-2.999)	-2.068x10 <sup>7</sup>	(-2.999)	-3.147x10 <sup>7</sup>	(-4.564)
$Q^r$	Pa (psi)	3.552x10 <sup>8</sup>	(51.518)	3.575x10 <sup>8</sup>	(51.853)	3.533x10 <sup>8</sup>	(51.241)	3.643x10 <sup>8</sup>	(52.839)	3.670x10 <sup>8</sup>	(53.243)	2.566x10 <sup>8</sup>	(37.219)	2.566x10 <sup>8</sup>	(37.220)	3.805x10 <sup>8</sup>	(56.630)
$Aa$ max	mm (in.)	.683 (.0273)	.856 (.0274)	.688 (.0271)	.741 (.0280)	.746 (.0282)	.591 (.0187)	.500 (.0187)	.500 (.0187)	.500 (.0187)	.500 (.0187)	.500 (.0187)	.500 (.0187)	.500 (.0187)	.500 (.0187)	.500 (.0187)	
$Ad$ max	mm (in.)	-.198 (-.0077)	-.198 (-.0078)	-.198 (-.0077)	-.201 (-.0078)	-.203 (-.0080)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	-.142 (-.0056)	
$E_{br}$	Nm (in.lb)	8.575 (75.896)	1.140 (10.092)	985.1 (8.219)	587.496 (5.288.287)	866.1 (78.452)	7.983.7 (70.662)	67.232.5 (695.058)	—	—	—	—	—	—	3.394 (30.043)	—	—
$E_f$	Nm (in.lb)	8.575 (75.896)	1.140 (10.092)	985.1 (8.219)	587.496 (5.288.287)	866.1 (78.452)	7.983.7 (70.662)	67.232.5 (695.058)	—	—	—	—	—	—	887.4 (7.854)	—	—
$E_c$	Mm (in.lb)	174.7 (5.546)	308.5 (2.713)	259.2 (2.294)	39.4 (349)	137.4 (1.216)	15.888.4 (138.854)	15.888.4 (138.854)	—	—	—	—	—	—	—	—	—
$P$	kg (lb)	70,693 (155.852)	75,751 (167.002)	70,706 (155.880)	81,987 (180.751)	86,672 (191.080)	23,833 (52.542)	23,833 (52.542)	23,833 (52.542)	23,833 (52.542)	23,841 (52.560)	153,476 (338.356)	—	—	—	—	—
$P^{cr}$	kg (lb)	70,693 (155.852)	75,751 (167.002)	70,706 (155.880)	81,987 (180.751)	86,672 (191.080)	23,833 (52.542)	23,833 (52.542)	23,833 (52.542)	23,833 (52.542)	23,841 (52.560)	153,476 (338.356)	—	—	—	—	—
$X_o$	cm (in.)	111.76 (44)	—	99.1 (39)	110.5 (43.5)	108.7 (42.8)	124.5 (49)	147.1 (57.9)	—	—	—	—	—	—	—	—	—
$X_{po}$	cm (in.)	114.3 (45)	88.9 (35)	25.1 (9.9)	23.1 (9.1)	22.4 (8.8)	149.9 (59)	152.4 (60)	—	—	—	—	—	—	—	—	—
$X_{max}$	cm (in.)	25.4 (10)	24.1 (9.5)	—	—	—	—	—	—	—	—	—	—	—	—	—	—
$X_{min}$	cm (in.)	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—
$V_e$	cm/sec (in./sec)	640 (252)	566.4 (223)	607.1 (239)	571.5 (225)	546 (215)	624.8 (246)	678.2 (267)	—	—	—	—	—	—	—	—	—
$V_i$	cm/sec (in./sec)	462.3 (182)	467.4 (184)	464.8 (183)	447 (176)	444.5 (175)	—	—	—	—	—	—	—	—	—	—	—
$V_m$	cm/sec (in./sec)	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—

\*For definitions of symbols in left-hand column see paragraphs 8, 10, or 11 as appropriate.

Another alternative is to add  $.1639 \text{ m}^3$  ( $10,000 \text{ in.}^3$ ) of fluid in a container connected to the recoil cylinder. As shown in tables 2 and 3, this would maintain positive pressures throughout the misfire cycle. However, the diameter of the recoil cylinder cannot be increased to accommodate the extra fluid without also increasing the wall thickness, which would change wall deflection, energy absorbing qualities, total weight, and probably most of the other parameters as well.

The maximum stresses in the cylinder wall are close to the pre-set limit of  $5.516 \times 10^8 \text{ Pa}$  ( $80,000 \text{ psi}$ ), which gives a static safety factor of about  $2\frac{1}{2}$ . The actual maximum stress is a tangential stress equal to  $6.0613 \times 10^8 \text{ Pa}$  ( $87,912 \text{ psi}$ ) and is acceptable. This assumes steel such as AISI 4130 or 4140, or their equivalent, is used throughout.

It was desired to keep the recoil cylinder long and narrow. This would give a low weapon profile and a "clean" efficient design. Since a cylinder longer than 2.44 or 2.74 meters (8 or 9 feet) is very difficult to machine, the effective length was set at 2.13 to 2.44 meters (7 to 8 feet).

The cylinder is 15.2 mm (0.60 inch) thick and is intended to deflect approximately .50 mm (0.0197 inch) (piston at latch) to absorb some of the energy.

The inner cylinder (sleeve) over which the piston slides is heavier (19.1 mm (0.75 inch) thick) to minimize deflection .142 mm (0.0056 inch) at latch). If it deflects too much, it could cling to the barrel which slides inside the sleeve and prevent the mechanism from functioning properly.

At one time, a multi-cylinder recoil system was considered. This would be a system with several recoil cylinders in a cluster around the barrel. After considering all the pros and cons, however, the idea of a multi-cylinder recoil system was discarded, mainly for the sake of simplicity and reliability.

## Dow Corning 200 Silicone Fluid

The volume and weight of the silicone fluid was finally reduced to .322 m<sup>3</sup> (85 gallons) 300.73 kg/663 pounds excluding the extra volume .1639 m<sup>3</sup> (10,000 in.<sup>3</sup>) which, if included, would total .488 m<sup>3</sup> (129 gallons), 457.67 kg (1,009 pounds). The volume of fluid displaced by the buffers .0339 m<sup>3</sup> (2,081 in.<sup>3</sup>) was not taken into account in the computer program since it was very uncertain initially what size and shape they would take. It would be advisable to design the buffers so that they occupy as little space as possible but still remain stiff. However, to compensate for the buffer volumes, which originally were assumed to be all fluid, an equivalent amount of fluid must be added to the cylinder volume. One way of doing this, would be to make the recoil cylinder 25.4 cm (10 inches) longer. This is shown in figure 5 as a 12.7 cm (5-inch) extension at each end of the 215.9 cm (85-inch) working length of the cylinder.

The silicon fluid is compressed only about 1.76% at latch during normal fire. The initial pressure at latch is  $2.068 \times 10^7$  Pa (3,000 psi) ( $P_1 = P_2 = P_3$ ). During cookoff, the compression reaches 2.85%. This is very low considering the compression could be 3 to 4% before the bulk modulus changes noticeably. Very high pressures in the recoil cylinder, however, would invalidate the linear relationship used in the mathematical model for the bulk modulus and make all the results erroneous.

The values used for the bulk modulus of the silicone fluid are all taken at room temperature. Since the bulk modulus is sensitive to temperature variations, this system would probably not work well in extreme climatic conditions without some modifications.

It has been suggested the effective bulk modulus be made softer or variable by incorporating a gas bag or spring-loaded expansion piston inside the recoil cylinder. The author, however, is opposed to adding anything to complicate the system unless it is absolutely necessary. (The system apparently works well in its present form.)

## Energy Stored

The computer was programmed to integrate the energies stored in the cylinder walls ( $E_c$ ), and in the fluid ( $E_f$ ), by a method of incremental summation. The values, however, which are listed in table 3, vary considerably from run to run and appear to be totally unreliable. Some of the values are only 1 to 2% of the breech energy. The only thing which these values clearly show is that  $E_f$  is many times larger than  $E_c$ . It is not the 50-50 relationship which was assumed at the beginning of the program. It would appear that the walls could be designed to store a larger percentage of the energy.

Knowledge of the actual amount of energy which is stored in the system is not essential, however. The only criterion for the feasibility of the system is that the recoiling mass is pushed back, after run-up, by the breech impulse to just beyond the latch position so there is positive latching every time. This is accomplished with the present design.

## Piston (Fig 5)

The piston is a long, thick cylinder (sleeve) with a 38.83 cm (14.5 inch) inside diameter and 9.53 mm (0.375 inch) thick walls. Its design is also the result of compromises. The ideal piston would be long and fairly thin, which would accommodate a long soft stroke with low pressures. However, with the piston too thin, its walls would crumble under the working pressures. Present wall thickness 9.53 mm (0.375 inch) is the minimum advised.

Calculations show that the critical buckling load of 5,203,792 kg (11,472,396 pounds) is far greater than the actual maximum load. The critical buckling load is questionable, however, since this is a very difficult value to determine at present (ref 3).

As a result of the piston being fairly thick, its working stroke is necessarily short. The maximum stroke is 64.26 cm (25.3 inch) for normal fire. The piston force, therefore, becomes fairly large. The maximum pressure occurring during cook-off surges to  $6.936 \times 10^7$  Pa (10,060 psi) which results in a rod pull of 153,476 kg (338,356 lb).

The average maximum piston load for normal fire is 77,111 kg (170,000 lb). These high loads are the result of trying to keep the weight of the oil down, and of a fairly thick-walled piston. However, the peak loads last only a fraction of a second.

The outer G-T seal "sees" a pressure of about  $3.172 \times 10^7$  Pa (4,600 psi) which is acceptable for this type of seal. (See appendix B.) The  $P_3$ , however, surges to  $6.936 \times 10^7$  Pa (10,060 psi) and the inside seal "sees" some of this pressure. This peak pressure, as mentioned earlier, is of a very short duration and only reaches the insnde seal through a long, tight squeeze under the piston and is consequently not considered critical.

The size and shape of the piston head is unimportant, except that it should have a sharp edge as shown in figure 5. This edge is part of the orifice and its sharpness eliminates much of the friction in the orifice, creating a more predictable flow coefficient. (See appendix C.)

Because of the relatively high initial pressures in the recoil cylinder and the relatively large effective piston area, the latch velocity is fairly high. It varied from 444.5 cm/sec (175 in./sec) to 482.6 cm/sec (190 in./sec). It appeared to be impossible to make this value any smaller without sacrificing some other elements of the design. Consequently, the latch mechanism must be capable of absorbing very large forces.

The firing or run-up velocities for normal fire varied from 546.1 cm/sec (215 in./sec) to 640 cm/sec (252 in./sec) in a run-up distance of 38.1 to 63.5 cm (15 to 25 in.). (This is at point of fire, after ignition delay or "coast" period.) There is, however, no real coast period since the pressure is always present and the velocity continues to increase. The fire control mechanism must start the firing operation at a lower velocity in anticipation of the desired firing velocity being reached. This will probably cause a problem since various ammunition has different time delays.

The run-up velocity, of course, must be adjustable for different zones of fire and different weapon elevations. Computer runs were made for 0°, 45°, and 75° elevations. The pressures, stresses, and velocities seemed to increase very little for the higher elevations and are very acceptable.

### Buffers (Fig 5)

The two buffers are identical, each having 50.8 cm (20 in.) effective buffing distance. The size of the buffer gap (annular orifice) was given much attention. It was recognized that the smaller this gap was at the beginning of buffing, the more effective the buffing would be in slowing down the mass. With a small gap, however, the tolerances and machining become very critical. This concerns gaps in the range of .254 mm (0.01 in.). Consequently, a gap of 1.27 mm (0.05 in.) which appeared satisfactory, was selected. Both buffers taper down from a gap of 1.27 mm (0.05 in.) to zero in 50.8 mm (20 in.). The outside of the buffers are also tapered. This, however, is for strength and rigidity. It is realized that a "give" of .254 mm (0.01 in.) in the buffers would be too much, and would disrupt the closely programmed buffing cycle.

There is one condition, however, when the front buffer seems inadequate. This is during misfire. It appeared that no matter how long the buffer was made, or how steep the taper, the piston would slow down but never quite stop before it would hit the bottom of the buffer pocket. To eliminate this condition, there was a suggestion to incorporate a stiff helical spring outside the front buffer. A 35.56 cm (14.) diameter by 15.24 cm (6 in.) long spring, which would compress about 3.81 cm (1.5 in.) was proposed. The author again is opposed to adding any kind of extra mechanism to disrupt the otherwise simple and "clean" design. As an alternative, the use of a "liquid spring" is suggested.

The elements are already present to make a liquid spring (see fig 5). All that is required is to make a straight bore at the end of the taper in the front buffer, and to shape the piston with a "flat" cylindrical surface to fit fairly tight into the bore like a typical piston. One would thus have a liquid spring.

Another solution would be to put an elastic pad (fig 5) inside the bottom of the front buffer as a stopper for the piston. This also seems feasible considering the velocity of the piston at this point is very low about 17.8 mm/sec (0.7 in./sec). A bumper pad of this type should probably be added in any case.

## Summary

In conclusion, the "undesirable" and the "desirable" features of the present, CFSR system will be repeated.

The undesirable points are:

1. Very high latch velocity.
2. Very high piston forces.
3. Piston hits bottom of front buffer.
4. Cavitation during some cycles.
5. Fairly large liquid volume and weight.

One feature which may be a disadvantage is that the recoil cylinder can be clamped only at its ends. The rest of the cylinder must be allowed to expand freely. Another feature which may be a disadvantage is the need to be able to "cock" the weapon in the field after a misfire. The piston must be pushed back to the latch position and the pressure raised to  $2.068 \times 10^7$  Pa (3,000 psi). As had been pointed out in the discussion, however, it is possible to design around most of these undesirable features.

The desirable features are (1) simple and reliable design with few parts, and (2) less maintenance required on the finished product. The author's unsubstantiated opinion is that the total recoil mechanism will be lighter in weight and also less costly to manufacture than a conventional recoil mechanism.

## RECOMMENDATIONS

The CFSR system appears to be a feasible and worthwhile concept to pursue further. It is recommended that a working model be built and tested based on the data given in this technical report. Some of the assumed parameters can then be verified or corrected as, for instance, the flow coefficient, the effective bulk modulus, stresses and strains in the metal parts, energy storing capability, and the actual seal friction. The mathematical model can then be corrected accordingly for reuse.

Dr. R. Yalamanchili has suggested making a CFSR mechanism with a cylindrical piston going right through the recoil cylinder. Although it is realized there would be some design problems, the author believes this is a very sound idea. This design would not have the undesirable features of the current concept.

The piston wall would be a little thicker where it enters the cylinder than where it exits, with a step in the thickness in between. The apparent advantage is in having a thick and stiff piston with a very small differential cross sectional area between the ends.

This design could have a long, soft stroke and low initial pressure. Less fluid volume (and weight) would be possible, cavitation could easily be eliminated, and there would be a much lower piston force and latch velocity.

Figure 6 shows this system as an alternate concept with approximate proportions and dimensions.

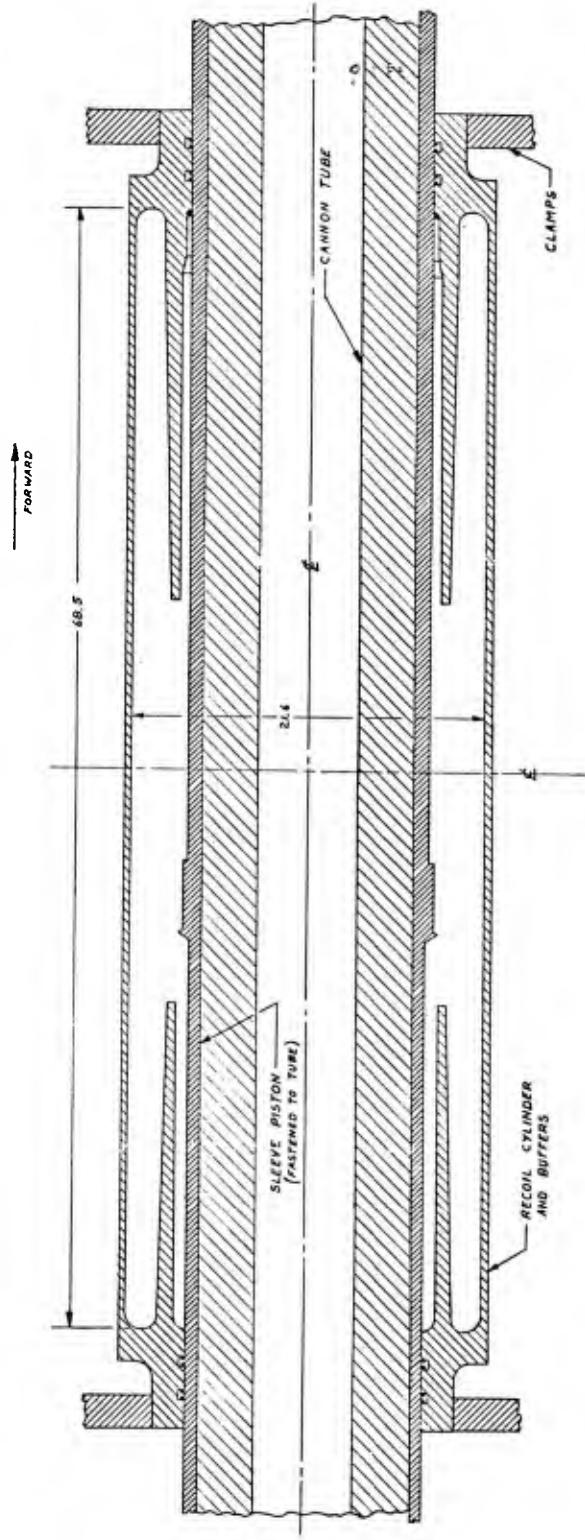


Figure 6. Preliminary CFSR concept study no. 2.  
(Dimensions are given in inches)

## REFERENCES

1. "155 MM Howitzer With Compressible Fluid/Soft Recoil (CFSR) Mechanism," Concept Proposal, GEN Thomas J. Rodman Laboratory, Rock Island Arsenal, Rock Island, IL, October 1975
2. Timoshenko, Stephen, Strength of Materials, Part II, Third Edition, D. Van Nostrand Co., Inc., NY, 1956
3. Flugge, Wilhelm, Stresses in Shells, Second Edition, Springer - Verlag, New York - Berlin, 1973

## BIBLIOGRAPHY

1. Nerdahl, M.C. and Frantz, J.W., "Modeling Effective Fluid Compressibility in a Puteaux Recoil Mechanism," Technical Report SWERR-TR-72-34, Artillery Weapon Systems Directorate, Weapons Laboratory, US Army Weapons Command, Rock Island, IL, June 1972
2. Seely, Fred B., and Smith, James O., Advanced Mechanics of Materials, John Wiley and Sons, Inc., NY, 1959
3. Military Handbook, MIL-HDBK-5, "Strength of Metal, Aircraft Elements," Armed Forces Supply Support Center, Washington, DC, March 1961
4. Daugherty, R.L. and Ingersoll, A.C., Fluid Mechanics With Engineering Applications, McGraw-Hill Book Co., Inc., NY, 1954
5. Hofgaard, Bjorn L., Technical Note R-TN-74-002, "Stress Analysis of 105 MM Towed Howitzer XM204 Serial Number 1 & 2 - Recoil Mechanism," GEN Thomas J. Rodman Laboratory, Rock Island Arsenal Rock Island, IL, January 1974

## APPENDIX A

### BREECH FORCE AND BREECH ENERGY

#### Breech Force

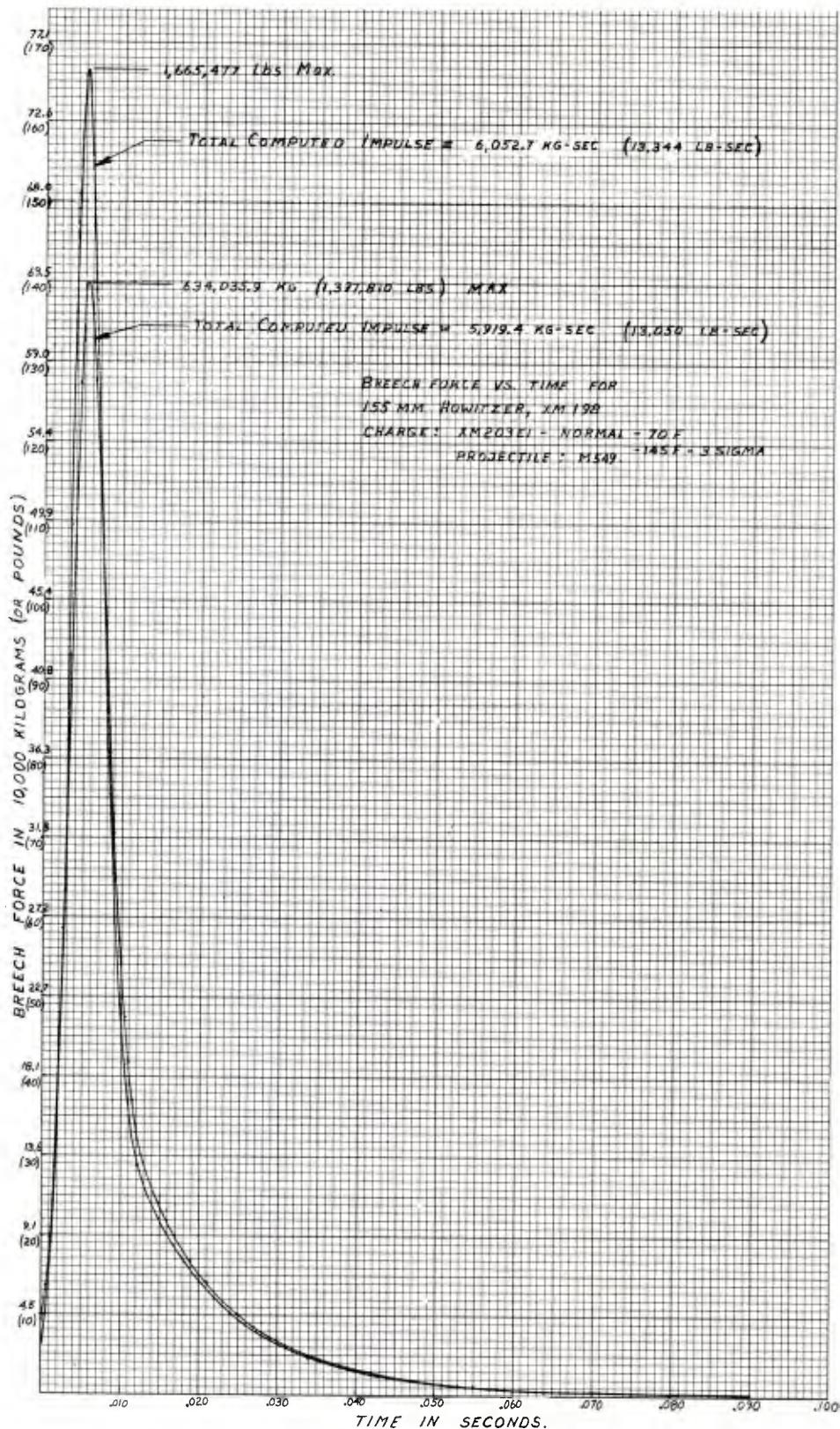
The breech force versus time curve is plotted from the M198 (155 mm) towed howitzer firing data. See page 42.

#### Breech Energy

The breech energy computations are given below

$$E_{br} = \frac{(I_{mp})^2 g}{2W_R} = \frac{(6052.737)^2 9.81}{2(2948.35)} = 60,948.58 \text{ meter-kg} \quad A1$$

$$E_{br} = \frac{(I_{mp})^2 g}{2W_R} = \frac{(13,344)^2 386.088}{2(6500)} = 5,288,287 \text{ in-lb} \quad A2$$

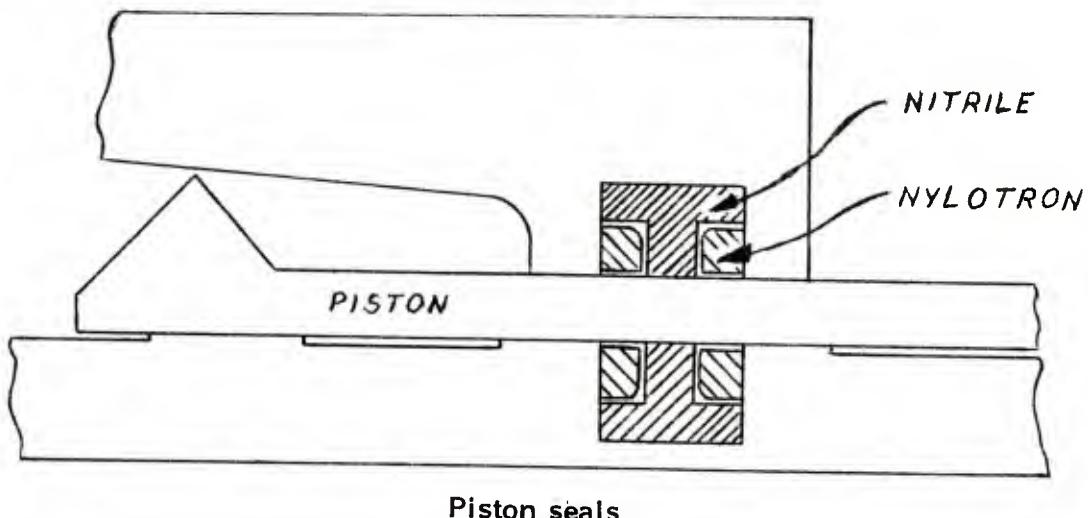


Breech force versus time

## APPENDIX B

### SEAL FRICTION

Two dynamic T-seals will probably be required, one outside and one inside the piston (see figure). The seals can be supplied by Greene, Tweed & Co., and are made from Nitrile or Buna-N (NBR) rubber with backup rings of nylotron.



Piston seals

An unresolved problem is that the coefficient of friction and the friction force are very uncertain for these materials in this application. The data available at this time does not permit an accurate, detailed analysis of oil pressure and seal friction. The friction formula given below should, however, be adequate for this study.

$$F = \mu_R v_R A_S P \quad (B-1)$$

where:

$F$  = Axial friction force on the piston

$A_S$  = Side area of seal backup ring plus edge of T-seal  
(Dimensions are taken from the Greene, tweed & Co.'s Palmetto catalog.)

$\mu_R$  = 0.50 = coefficient of friction (An approximate figure for rubber supplied by RIA Rubber Laboratory.)

$\nu_R$  = 0.48 = Poisson's ratio for rubber (An approximate figure supplied by the RIA Rubber Laboratory.)

P = Fluid pressure in psi.

Inside seal:

$$*A_{S_1} = \pi(7.25^2 - 7.00^2) = 11.1919 \text{ in}^2 \quad (\text{B-2})$$

$$*F = 0.50(.48)11.1919P = 2.69 P. \quad (\text{B-3})$$

Outside seal:

$$*A_{S_2} = \pi(8.00^2 - 7.75^2) = 12.37 \text{ in}^2 \quad (\text{B-4})$$

$$*F = 0.50(.48)12.37P = 2.97 P. \quad (\text{B-5})$$

This relationship, variable with pressure, was used in the mathematical model. It is the best approximation available for the seal friction at the present time.

Since the piston will be machined to slide over the inside sleeve and probably will have a fairly close sliding fit in certain areas, it is believed that the inside seal never "sees" the full peak pressure which only lasts a fraction of a second. It was therefore arbitrarily decided to use only 50% of the force on this seal, which accounts for the 0.5 factor in equation 79.

Greene, Tweed and Co., North Wales, PA can supply G-T Ring seals for pressures from zero and up to  $6.895 \times 10^7$  Pa (10,000 psi) and higher. The computed CFSR pressures are all well within this range.

The final dimensions of the piston and seals are not the exact dimensions used in the friction calculations shown above and used in the mathematical model. There should not, however, be any significant change in the forces.

---

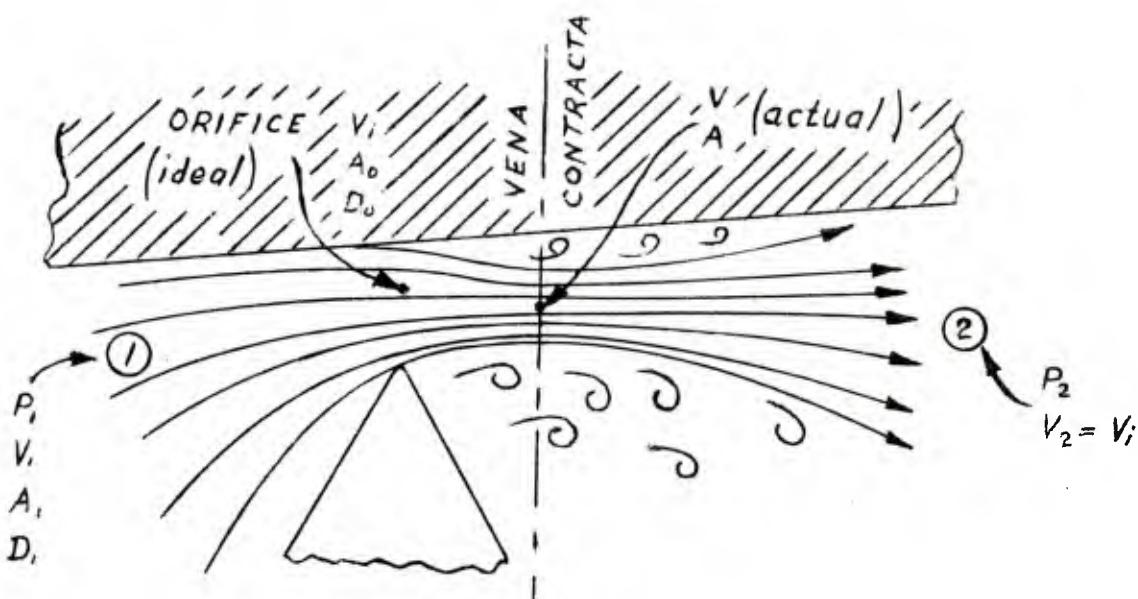
\*These formulas are given in English units.

## APPENDIX C

### FLOW COEFFICIENT

The following information can be found in any textbook on fluid mechanics, but is included here to emphasize the number of physical conditions which the flow coefficient is dependent upon.

Looking at the flow going through an orifice (see figure), some coefficients and relationships are established.



### Flow through orifice

The relationship between actual velocity ( $V$ ) and ideal frictionless velocity ( $V_i$ ) at "vena contracta" is determined by the coefficient of velocity ( $C_v$ ), thus:

$$V = C_v V_i \quad (C-1)$$

Vena contracta is the minimum cross section of the jet where it is contracted right outside the orifice on the down-stream side. This contraction is dependent on friction and shape of the orifice and the viscosity of the fluid.

The ratio of the cross-sectional area of a jet ( $A$ ) at vena contracta to the area at the orifice ( $A_o$ ) is called the coefficient of contraction ( $C_c$ ). This relationship can be expressed as follows:

$$A = C_c A_o . \quad (C-2)$$

The ratio of the actual rate of discharge ( $Q$ ) to the ideal rate of discharge ( $Q_i$ ), if there were no friction and no contraction, may be defined as the coefficient of discharge ( $C_d$ ).

Thus:

$$Q = C_d Q_i \quad (C-3)$$

By observing that  $Q = AV$  and  $Q_i = A_o V_i$ , it is seen that:

$$C_d = C_c C_v . \quad (C-4)$$

Writing the Bernoulli's equation thru the orifice in the buffer, it

$$\frac{P_1}{\omega} + \frac{V_1^2}{2g} = \frac{P_2}{\omega} + \frac{V_2^2}{2g} \quad (C-5)$$

(See figure.)

Rearranging this equation to get the flow velocity outside the orifice ( $V_2 = V_i$ ), it becomes

$$V_2 = \sqrt{\frac{2g}{\omega} (P_1 - P_2) + V_1^2} \quad (C-6)$$

and the actual jet velocity is

$$V = C_v \sqrt{\frac{2g}{\omega} (P_1 - P_2) + V_1^2} . \quad (C-7)$$

By applying the continuity equation

$$V_1 A_1 = AV \quad (C-8)$$

$$V_1 = AV/A \quad (C-9)$$

$$V_1 = C_c A_0 V/A \quad (C-10)$$

$$V_1 = C_c V (D_0/D_1)^0 \quad (C-11)$$

and substituting this  $V_1$  value into the Bernoulli equation, it becomes

$$V = \frac{C_v}{\sqrt{1 - C_d^2 (D_0/D_1)^4}} \sqrt{\frac{2g}{\omega} (P_1 - P_2)} \quad (C-12)$$

Since

$$Q = C_c A_0 V, \quad (C-13)$$

$$Q = \frac{C_d A_0}{\sqrt{1 - C_d^2 (D_0/D_1)^4}} \sqrt{\frac{2g}{\omega} (P_1 - P_2)} \quad (C-14)$$

Now to simplify this equation,

$$\sqrt{\frac{C_d}{1 - C_d^2 (D_0/D_1)^4}} = K \quad (C-15)$$

$K$  is called the flow coefficient, and the final equation becomes

$$Q = KA_0 \sqrt{\frac{2g}{\omega} (P_1 - P_2)} \quad (C-16)$$

This is the equation which is used in the mathematical model.

Since there is a variable orifice-- $D_o$ ,  $D_1$ , and  $A_o$  vary--  $K$  is not a constant. In addition,  $C_d$ ,  $C_v$ , and  $C_c$ , which are dependent on orifice shape and friction or drag and viscosity, vary. Viscosity of the fluid is dependent on pressure and temperature. It must also be remembered that the specific weight of the fluid  $\omega$  varies with pressure and temperature.

Unfortunately, there are no reliable flow coefficients available for this fluid. It was assumed, however, that there is an average constant coefficient for this application which would apply for the given conditions and boundaries, accounting for all the variables including  $\omega$ . The very best estimate for such a value is,

$$K = 0.95 \quad (C-17)$$

## APPENDIX D

### BULK MODULUS

Dow Corning 200 (10 cs), a silicon fluid, is used as the working fluid in the recoil mechanism. This liquid has good compressibility and stability and a relatively flat viscosity curve.

However, many of its physical properties are not very well known. There are no precise data available either for the absolute bulk modulus, or for the effective bulk modulus which includes the container--in this case the recoil cylinder. The bulk modulus ( $\beta$ ), an essential parameter for the solution of this recoil problem, varies with pressure and temperature over a wide range.

The bulk modulus is defined by the following differential equation:

$$\beta = - V_i \frac{dP}{dv} , \quad (D-1)$$

where

$P$  = Fluid pressure

$V_i$  = Initial fluid volume

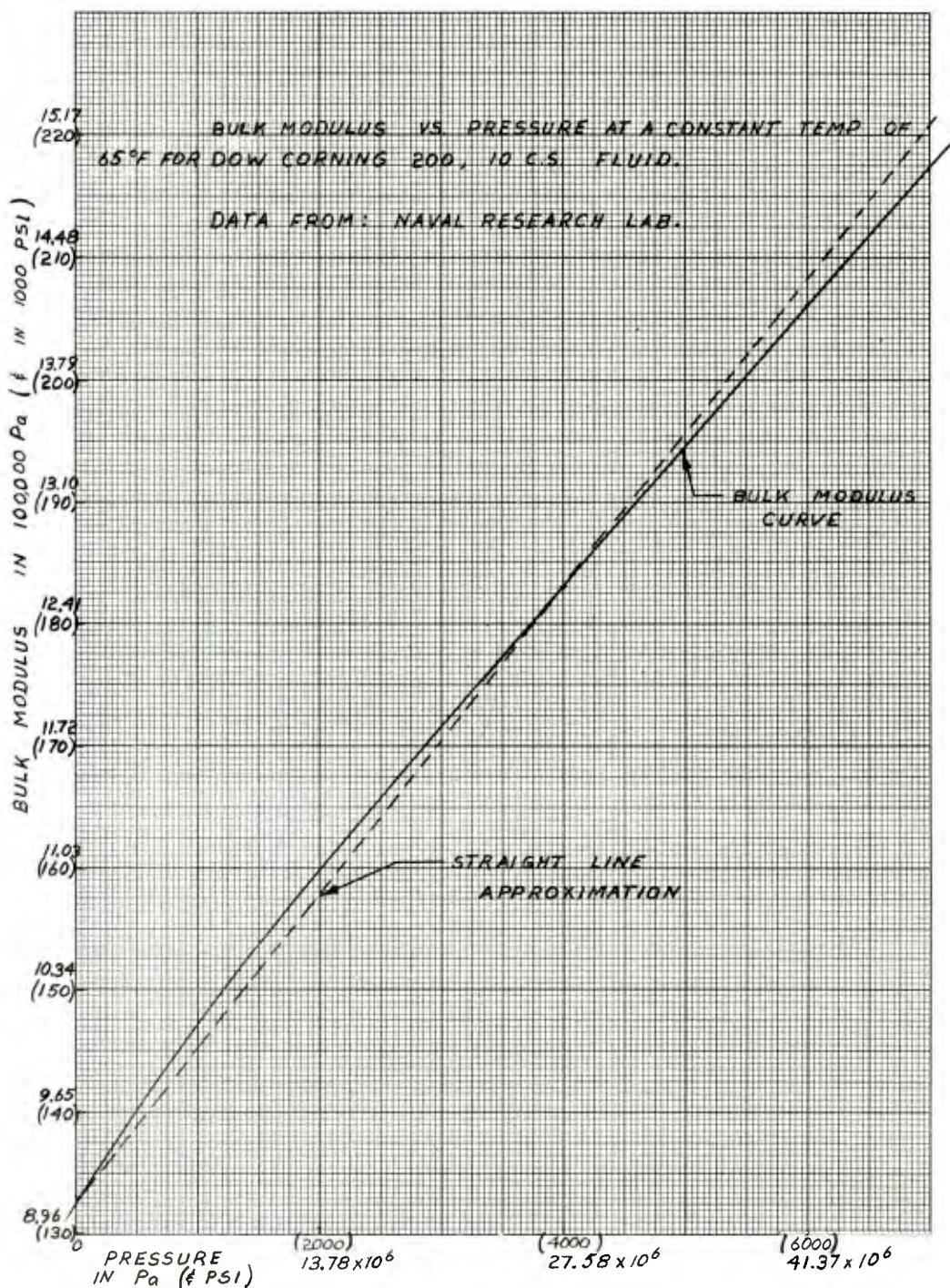
$V$  = Fluid volume at time ( $t$ ) .

The compressibility ( $k$ ) of the fluid is defined as

$$k = \frac{1}{\beta} , \quad (D-2)$$

A few values of bulk modulus for four different pressures, covering approximately half of the pressure range of the recoil mechanism, were published by the Naval Research Laboratory. Since pressures in a recoil mechanism vary greatly during a cycle, and temperatures vary only a small amount, a curve was plotted for room temperature conditions only (65°F), using the four points available. This curve proved to be practically a straight sloping line, so a straight line was used for the needed  $\beta$  versus  $P$  relationship (see figure). The tangent to this curve is 12.67. Thus, the following relationship was used in the mathematical model:

$$\begin{aligned} \beta &= 12.67 P + 9.136 \times 10^8 \\ (\beta &= 12.67 P + 132,500) . \end{aligned} \quad (D-3)$$



Bulk modulus versus pressure

**APPENDIX E**

**COMPUTER PRINTOUTS**  
**(TOTAL PROGRAM)**

\*DARCOM MIDWEST SECC\* RM9.PRI A..START JOB 7994...2.03.21 PM 15 JUN 77.2T03.K173.KW3PDB70.BENZKOFFER  
\*DARCOM MIDWEST SECC\* RM9.PRI A..START JOB 7994...2.03.21 PM 15 JUN 77.2T03.K173.KW3PDB70.BENZKOFFER  
\*DARCOM MIDWEST SECC\* RM9.PRI A..START JOB 7994...2.03.21 PM 15 JUN 77.2T03.K173.KW3PDB70.BENZKOFFER

\*DARCOM MIDWEST SECC\*  
\*DARCOM MIDWEST SECC\*  
\*DARCOM MIDWEST SECC\*

H A S P   J O B   L O G

\$13.51.19 JOB 7994 - KW3PDB70 - RDR ON 13.22.18 - BEGINNING EXEC - INIT 4 - CLASS A  
13.51.21 JOB 7994 IEF403I KW3PDB70 STARTED TIME=13.51.21  
\$13.52.20 JOB 7994 ESTIMATED TIME EXCEEDED  
13.57.46 JOB 7994 IEF404I KW3PDB70 ENDED TIME=13.57.46  
N13.57.47 JOB 7994 END EXECUTION.

----- HASP-II JOB STATISTICS -----

426 CARDS READ

1.214 SYSOUT PRINT RECORDS

0 SYSOUT PUNCH RECORDS

6.49 MINUTES EXECUTION TIME

```

//KW3PDB70 JOB (2T03,K173.1.3.9999,000065),'BENZKOFFR'
// EXEC FORTGCLG,REGION=150K
XXDEFAULTS PROC SYSOUT=A
XXFORT EXEC PGM=IEYFORT,REGION=106K,DRTRY=(3,3)
XXSYSPRINT DD SYSOUT=&SYSOUT
IEF653I SUBSTITUTION JCL - SYSOUT=A
XXSYSLIN DD DSN=&OBJEKT,UNIT=TEL,SPACE=(CYL,(1,1)),
          DCB=BLKSIZE=800,DISP=(,PASS),
/FORT.SYSIN DD *
IEF236I ALLOC. FOR KW3PDB70 FORT
IEF237I OC2 ALLOCATED TO SYSPRINT
IEF237I 114 ALLOCATED TO SYSLIN
IEF237I 090 ALLOCATED TO SYSIN
IEF142I - STEP WAS EXECUTED - COND CODE 0000
IEF285I SYS77165.T131819.RV000.KW3PDB70.OBJECT
IEF285I VOL SER NOS= IFSEOS.
IEF373I STEP /FORT / START 77166.1351
IEF374I STEP /FORT / STOP 77166.1351 CPU 0MIN 16.82SEC MAIN 100K CC= 0
      *****
*STEP FORT *JOB KW3PDB70 *****
*RESOURCE- CORE(K) DISK(10) TAPE(10)--UNITS(U) IN-HASP(I0)-OUT OTHER(I0) CPU TIME(C) STEP TIME(T)
*USAGE- 150 53 0 0 404 553 0 0010016.82 00100125.87
*****



XXLINKED EXEC PGM=IEWLF800,PARM='LIST,LET,XREF+REGION=100K,
COND=(4,LT,FORT),DRTRY=(3,3)
XX
 4 XXSYSPRINT DD SYSOUT=&SYSOUT
IEF653I SUBSTITUTION JCL - SYSOUT=A
XXSYSLIB DD DSN=NONG.FORTLIB,DISP=SHR
XXSYSLIN DD DSN=&OBJEKT,DISP=(OLD,DELETE)
XX
XXSYSUT1 DD DNAME=SYSIN
XXSYSLMD DD DSN=&LOAD(MAIN),UNIT=TEL,SPACE=(CYL,(1,1))
XX
DISP=(,PASS)
IEF236I ALLOC. FOR KW3PDR70 LKED
IEF237I OC2 ALLOCATED TO SYSPRINT
IEF237I 115 ALLOCATED TO SYSLIB
IEF237I 114 ALLOCATED TO SYSLIN
IEF237I 114 ALLOCATED TO SYSUT1
IEF237I 114 ALLOCATED TO SYSLMD
IEF142I - STEP WAS EXECUTED - COND CODE 0000
IEF285I SYS1.FORTLIB
IEF285I VOL SER NOS= IFSEPP.
IEF285I SYS77165.T131819.RV000.KW3PDB70.OBJECT
IEF285I VOL SER NOS= IFSEOS.
IEF285I SYS77165.T131819.RV000.KW3PDB70.R0004992
IEF285I VOL SER NOS= IFSEOS.
IEF285I SYS77165.T131819.RV000.KW3PDB70.LOAD
IEF285I VOL SER NOS= IFSEOS.
IEF373I STEP /LKED / START 77166.1351
IEF374I STEP /LKED / STOP 77166.1352 CPU 0MIN 02.24SEC MAIN 98K CC= 0
      *****
*STEP LKED *JOB KW3PDB70 *****
*RESOURCE- CORE(K) DISK(10) TAPE(10)--UNITS(U) IN-HASP(I0)-OUT OTHER(I0) CPU TIME(C) STEP TIME(T)
*USAGE- 150 337 0 0 163 0 001002.24 0010015.57
*****
```

```

XXGO      EXEC PGM=*,LKED,SYSMOD,COND=((4,LT,FORT),(4,LT,LKED)),    00001500
          DPRTY=(3,3)                                         00001600
XXFT05F001 DD  DDNAME=SYSIN                                00001700
XXFT06F001 DD  SYSOUT=SYSOUT                                00001800
IEF653I SUBSTITUTION JCL - SYSOUT=A
XXFT07F001 DD  SYSOUT=B
//GO.SYSIN DD *
//
```

```

IEF236I ALLOC. FOR KW3PDB70 GO
IEF237I 114 ALLOCATED TO PGH=+DD
IEF237I 090 ALLOCATED TO FT05F001
IEF237I 0C2 ALLOCATED TO FT06F001
IEF237I 1A2 ALLOCATED TO FT07F001
IEF142I - STEP WAS EXECUTED - COND CODE 0000
IEF285I SYS77165.T131819.RV000.KW3PDB70.LOAD
IEF285I VOL SER NOS= IFSEOS.
IEF373I STEP /GO / START 77166.1352
IEF374I STEP /GO / STOP 77166.1357 CPU 3MIN 19.59SEC MAIN 48K CC= 0
*STEP GO **JOB KW3PDB70*****
*RESOURCE= CORE(1K) DISK(10) TAPE(10)--UNITS(IU) IN-HASP(I/O)-OUT OTHER(I/O) CPU TIME(C) STEP TIME(T)
*USAGE- 150 0 0 20 0 0010319.59 00105134.17 *
*****DELETED*****
IEF285I SYS77165.T131819.RV000.KW3PDB70.LOAD
IEF285I VOL SER NOS= IFSEOS.
IEF375I JOB /KW3PDB70/ START 77166.1351
IEF376I JOB /KW3PDB70/ STOP 77166.1357 CPU 3MIN 38.65SEC
*****DELETED*****
*JOB KW3PDB70**2103**K173**RENZKOFER*****
*UNITS= CORE(K*T) DISK(10) TAPE(I/O) TAPE(U*T) HASP(I/O) OTHER(I/O) CPU(C) TOTAL
*COSTS- $1.97 $0.12 $0.00 $0.00 $0.00 $13.36 $15.45
*****
```

```

REAL L1,L2,L3,L4,MR,N,NU,I,J
EXTERNAL F7,F8,F9
DIMENSION TIME(81),BFORCE(81)
COMMON/BLK1/H,HH
COMMON/BLK2/DI,F,HI,I,J,R,S,TI,U
COMMON/BLK3/IP,PB1,PB2,PB3
COMMON/BLK4/A,R,C,D,G,CD,RHOE,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA
COMMON/BLK5/A1,A2,A3,A4,A5,R1,R2,R3,R4,RS
NAMELIST/NAM/R1,R2,R3,R4,R5,A,B,C,D,G,CD,RHOF,RHOS,L1,L2,L3,
1E,N,NU,V0,WR,GAMMA
NAMELIST/NAM1/TIME,BFORCE
READ(15,NAM)
WRITE(6,NAM)
READ(5,NAM1)
WRITE(6,NAM1)
MR=WR/G
DI=0.
U=0.
F=0.
R=1.
H1=0.
J=0.
TI=-1.
S=-1.
I=0.
IMISFR=0
INORM=1
IRBUFF=0
VE=240.
ICHECK=0
IPRESS=0
T=0.
PB1=0.
PB2=0.
PB3=0.
P2=3000.
P1=P2
P3=30000.
L4=L1+L2+L3
L4=85.
DELA=3.1416*L4*P2*A/(3.1416*L4*E/(B**2*(1.-NU)*A**2)/(B**2-
1A**2)+NU)
DELD=-3.1416*L4*P2*D/(3.1416*L4*E/(D**2*(1.-NU)*C**2)/(D**2-
1C**2)-NU))
DELA0=0.
DELD0=0.
A1=3.1416*(R2**2-R3**2)
A2=3.*416*(R1**2-R3**2)
A3=3.*1416*(R3**2-R5**2)
A4=3.1416*(R3**2-R4**2)
A5=3.1416*(R4**2-R5**2)
X=L3
XD=0.

```

C

FORTRAN IV G LEVEL 21

MAIN DATE = 77166

13 / 51 / 23

SAGE 0003

```

0050 HF=000005
0051 HH=H/2.
0052 SUM=L2+L3
0053 IP=1
0054 KOUNT=0
0055 EEF=0.
0056 EC=0.
0057 IF (INORM.EQ.1) GO TO 3
0058 IF (INORM.EQ.0.AND.IMISFR.EQ.1) GO TO 12
0059 1 CALL LINEAR(T,TIME,BFORCE,PB1)
0060 T=T+HH
0061 CALL LINEAR(T,TIME,BFORCE,PB2)
0062 T=T+HH
0063 CALL LINEAR(T,TIME,BFORCE,PB3)
0064 IP=1
0065 3 CALL KUTTA(X,XD,P3,P2)
0066 XDD=F1(X,XD,P3,P2)
0067 CALL KUTTA2(DELA,DELD,DELAD,DELDD,P2)
0068 DELADD=FS(DELA,DELAD,P2)
0069 DELDD=F6(DELD,DELDD,P2)
0070 CALL KUTTA1(X,XD,P2,P3,DELA,DELAD,DELDD)
0071 P1D=F2(X,XD,P1,P2)
0072 P2D=F3(X,XD,P1,P2,P3,DELA,DELAD,DELDD)
0073 P3D=F4(X,XD,P2,P3)
0074 IF (X.LT.SUM) GO TO 13
0075 DI=1.
0076 U=1.
0077 F=1.
0078 GO TO 13
0079 12 T=T+H
0080 CALL KUTTA(X,XD,P3,P1)
0081 XDD=F5(DELA,DELAD,P2)
0082 IF (X.LT.SUM) GO TO 16
0083 DI=1.
0084 U=1.
0085 F=1.
0086 16 CALL KUTTA2(DELA,DELD,DELAD,DELDD,P2)
0087 DELADD=FS(DELA,DELAD,P2)
0088 DELDD=F6(DELD,DELDD,P2)
0089 CALL KUTTA1(X,XD,P2,P3,DELA,DELAD,DELDD)
0090 P1D=F2(X,XD,P1,P2)
0091 P2D=F3(X,XD,P1,P2,P3,DELA,DELAD,DELDD)
0092 P3D=F4(X,XD,P2,P3)
0093 IF (X.GT.SUM) P3=P2
0094 IF (X.GT.SUM) GO TO 14
0095 P1=P2
0096 P3=P2
0097 DI=0.
0098 U=0.
0099 F=0.
0100 GO TO 14
0101 13 IF (IRBUFF.EQ.1) GO TO 11
0102 IF (X.GT.SUM) GO TO 19

```

```

0103      DI=0.
0104      U=0.
0105      F=0.
0106      P1=P2
0107      P3=P2
0108      P1=P2
0109      GO TO 14
0110
0111      P3=P2
0112      SIGT=(R**2+A**2)/(B**2-A**2)*P2
0113      SIGAX=A**2*P2/(B**2-A**2)
0114      SIGR=SQRT((SIGT-SIGR)**2+(SIGR-SIGAX)**2*(SIGAX-SIGT)**2)
0115      V1=3.*1416*(A**2-D**2)*L4+V0
0116      KOUNT=KOUNT+1
0117      IF (P2.LE.0.) GO TO 20
0118      V=3.*1416*(A*DELA)**2-(D*DEL0)**2*L4+A5*(L4-X)
0119      EF=EF-F9(N,P2,P2D,N)*H
0120      EC=EC+2.*3.1416*L4*(F7(A*DELA,DELAD,P2)+FB(D,DEL0,DELDD,P2))*H
0121      IF (KOUNT.LT.40) GO TO 7
0122      KOUNT=0
0123      PRINT 5,T,X,XD,XDD,P1,P1D,P2,P2D,P3,P3D
0124      5 FORMAT(2X,'T=''',F6.4,1X,'X=''',F6.3,1X,'XD=''',F9.2,1X,'XDD=''',F9.1,1X,
1*P1=''',F9.2,1X,'P1D=''',F10.1,1X,'P2=''',F9.2,1X,'P2D=''',F10.1,1X,
2*P3=''',F9.2,1X,'P3D=''',F9.1)
0125      PRINT 2*DFLA,DELA,DEL0,DELDD,SIGT,SIGAX,SIGR,SIGE
0126      2 FORMAT(2X,DELA,'F7.4,1X,'SIGT=''',F7.4,1X,'DEL0=''',F7.4,1X,
1*DELDD=''',F7.4,1X,'SIGT=''',F10.1,1X,'DEL0=''',F7.4,1X,
2*SIGR=''',F10.1,1X,'SIGE=''',F10.1)
0127      PRINT 2*DELA,DELD,DEL0,DELDD,SIGT,SIGAX,SIGR,SIGE,EC,EF
0128      2 FORMAT(2X,'DELA=''',F7.4,1X,'DELAD=''',F7.4,1X,'DEL0=''',F7.4,1X,
1*SIGT=''',F10.1,1X,'SIGAX=''',F10.1,1X,'SIGR=''',F10.1,1X,
2*SIGR=''',F10.1,1X,'SIGE=''',F10.1)
0129      PRINT 2*DELA,DELD,DEL0,DELDD,SIGT,SIGAX,SIGR,SIGE,EC,EF
0130      2 FORMAT(2X,'DELA=''',F7.4,1X,'DEL0=''',F7.4,1X,
1*SIGT=''',F10.1,1X,'SIGAX=''',F10.1,1X,
2*SIGR=''',F10.1,1X,'SIGE=''',F10.1,1X,'SIGF=''',F15.1,1X,'EFF=''',F15.1)
0131      IF (XDD.LE.0..OR.XD.GE.YE) GO TO 4
0132      T=T+H
0133      4 ICHECK=ICHECK+1
0134      IF (ICHECK.EQ.1) TT=0.
0135      CALL LINEAR(TT,TIME,BFORCE,PB1)
0136      TT=TT+HH
0137      CALL LINEAR(TT,TIME,BFORCE,PB2)
0138      TT=TT+HH
0139      CALL LINEAR(TT,TIME,BFORCE,PB3)
0140      IP=1
0141      T=T+H
0142      HI=1.
0143      IF (XD.GE.0..) GC TO 8

```

FORTRAN IV G LEVEL 21

MAIN DATE = 77166

13/51/23

PAGE 0004

```
0144      S=1.
          60 T0 9
0145      8
          S=-1.
0146      TI=-1.
0147      9 IF(X.GT.L3.AND.IRBUFF.EQ.0) GO T0 3
0148      I=-1.
0149      J=1.
0150      R=0.
0151      0152      TI=1.
0153      IRBUFF=1
          IF(XD.GT.0..AND.X.GT.L3) GO TO 6
0154      GO TO 3
0155      10 IF(XD.LT.0..AND.IPRESS.EQ.1.AND.X.LT.L3) GO TO 1
0156      IF(XD.LT.0..AND.IPRESS.EQ.1.AND.X.GE.L3) GO TO 18
0157      IF(XD.LT.0..AND.IPRESS.EQ.0.AND.X.GE.L3) GO TO 18
0158      IF(XD.GT.0..AND.IPRESS.EQ.0.AND.X.GT.L3) GO TO 17
0159      IF(XD.GT.0..AND.IPRESS.EQ.0.AND.X.GE.L3) GO TO 18
0160      IF(XD.GT.0..AND.IPRESS.EQ.1.AND.X.LT.L3) GO TO 1
0161      IF(XD.GT.0..AND.IPRESS.EQ.1.AND.X.GE.L3) GO TO 6
0162      17 IPRESS=1
0163      GO TO 1
0164      18 X=L3
0165      XD=0.
0166      GO TO 1
0167      15 IF(XD.GT.0.) GO TO 12
0168      6 PRINT 21,EC,EF
0169      21 FORMAT(5X,'EC='',F30.5,5X,'EF='',F30.5)
0170      CALL EXIT
0171
```

59

FORTRAN IV G LEVEL 21

MAIN DATE = 77166

13/51/23

PAGE 0005

```
*OPTIONS IN EFFECT* NOID,ERCDIC,SOURCE,NOLIST,NODECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME = MAIN      * LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 171,PROGRAM SIZE = 5500
*STATISTICS* NO DIAGNOSTICS GENERATED
```

FORTRAN IV G LEVEL 21 AF DATE = 77166 13/51/23 PAGE 0001

```
0001      FUNCTION AF(X)
0002      REAL L1*L2*L3,L4*MR*N*NU*I*J
0003      COMMON/BLK4/A,B,C,D,G,CD,RHOF,RHOS,L1*L2*L3,L4,E,N,NU,V0,MR,GAMMA
0004      COMMON/BLK5/A1,A2,A3,A4,A5,R1,R2,R3,R4,R5
0005      GF=R2-R3
0006      AF=A1+3.1416*GF**2*(L2*L3)**2/L1**2*2.*3.*1416*GF*R2*(L2*L3)/L1-
1(2.*3.*1416*GF**2*(L2*L3)/L1**2*2.*3.*1416*GF*R2/L1)*X*
23.1416*GF**2*X**2/L1**2
0007      RETURN
0008      END
```

FORTRAN IV G LEVEL 21 AF DATE = 77166 13/51/23 PAGE 0002

```
*OPTIONS IN EFFECT* NOID,ERCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME = AF LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = R,PROGRAM SIZE = 484
*STATISTICS* NO DIAGNOSTICS GENERATED
```

FORTRAN IV G LEVEL 21 AR DATE = 77166 13/51/23 PAGE 0001

```
0001      FUNCTION AR(X)
0002      REAL L1*L2*L3*L4,MR,N,NU,I,J
0003      COMMON/BLK4/A,B,C,D,G,CD,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA
0004      COMMON/BLK5/A1,A2,A3,A4,A5,R1,R2,R3,R4,RS
0005      GR=R1-P3
0006      AR=2.*3.1416*GR*R3*X/L3+3.1416*GR**2*X**2/L3**2
0007      RETURN
0008      END
```

FORTRAN IV G LEVEL 21 AR DATE = 77166 13/51/23 PAGE 0002

```
*OPTIONS IN EFFECT* NOID=ERCDIC, SOURCE=NOLIST, NODECK, LOAD, NOMAP
*OPTIONS IN EFFECT* NAME = AR   LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 8, PROGRAM SIZE = 380
*STATISTICS* NO DIAGNOSTICS GENERATED
```

FORTRAN IV G LEVEL 21 QF DATE = 77166 13/51/23  
PAGE 0001

0001 FUNCTION QF(P1,P2,X)  
0002 REAL L1,L2,L3,L4,MR,N,NU,T,J  
0003 COMMON/BLK4/A,B,C,D,G,CD,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA  
0004 QF=AF(X)\*CD\*SQRT(2.\*G\*ABS(P1-P2)/RHOF)\*SGN(P1-P2)  
0005 RETURN  
0006 END

FORTRAN IV G LEVEL 21 QF DATE = 77166 13/51/23  
PAGE 0002

\*OPTIONS IN EFFECT\* NOID,FBCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP  
\*OPTIONS IN EFFECT\* NAME = QF LINECNT = 55  
\*STATISTICS\* SOURCE STATEMENTS = 6\*PROGRAM SIZE = 478  
\*STATISTICS\* NO DIAGNOSTICS GENERATED

FORTRAN IV G LEVEL 21

DATE = 77166

PAGE 0001

```
0001      FUNCTION QR(P2,P3,X)
0002      REAL L1,L2,L3,L4,MR,N,NU,I,J
0003      COMMON/BLK4/A,B,C,D,G,CD,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA
0004      QR=AR(X)*CD*SQRT(2.*G*ABS(P3-P2)/RHOF)*SGN(P3-P2)
0005      RETURN
0006      END
```

FORTRAN IV G LEVEL 21

DATE = 77166

PAGE 0002

```
*OPTIONS IN EFFECT* NOID,ERCHIC,SOURCE,NOLIST,NODECK,LOAD,NDMAR
*OPTIONS IN EFFECT* NAME = QR   LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 6,PROGRAM SIZE = 478
*STATISTICS* NO DIAGNOSTICS GENERATED
```

FORTRAN IV G LEVEL 21 F1 DATE = 77166 PAGE 0001  
 13/51/23  
 0001 FUNCTION F1(X,XD,P3,P1)  
 0002 REAL L1,L2,L3,L4,MR,N,NU,J,J  
 0003 COMMON/BLK2/DI,F,HI,I,J,R,S,T1,U  
 0004 COMMON/BLK4/A,B,C,D,G,CD,RHDF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA  
 0005 COMMON/BLK5/A1,A2,A3,A4,A5,R1,R2,R3,R4,RS  
 0006 IF(XD.GT.0.) S=1.  
 0007 IF(XD.LE.0.) S=1.  
 0008 F1=1./MR\*(S\*(2.69\*.5\*P3+2.97\*P1)+P3\*A3-P1\*A4-HI\*PBF(XXX)-  
 1\*MR\*G\*SIN(GAMMA))  
 0009 RETURN  
 0010 END

FORTRAN IV G LEVEL 21 F1 DATE = 77166 PAGE 0002  
 13/51/23  
 \*OPTIONS IN EFFECT\* NOID,EBCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP  
 \*OPTIONS IN EFFECT\* NAME = F1 , LINECNT = 55  
 \*STATISTICS\* SOURCE STATEMENTS = 10\*PROGRAM SIZE = 618  
 \*STATISTICS\* NO DIAGNOSTICS GENERATED

FORTRAN IV G LEVEL 21 F2 DATE = 77166 13/51/23 PAGE 0001  
 0001 FUNCTION F2(X,XD,P1,P2)  
 0002 REAL L1,L2,L3,L4,MR,N,NU,I,J  
 0003 COMMON/BLK2/D1,F,HI,I,J,R,S,T,U  
 0004 COMMON/BLK4/A,B,C,D,G,CD,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA  
 0005 COMMON/BLK5/A1,A2,A3,A4,A5,R1,R2,R3,R4,RS  
 0006 F2=U\*(N\*P1+132500.)\*(A4\*XD-QF(P1,P2,X))/((A4+AF(X))\*((A4+AF(X))+(L1\*L2\*L3-X))  
 0007 RETURN  
 0008 END

FORTRAN IV G LEVEL 21 F2 DATE = 77166 13/51/23 PAGE 0002  
 \*OPTIONS IN EFFECT\* NOID, EBCDIC, SOURCE, NOLIST, NODECK, LOAD, NOMAP  
 \*OPTIONS IN EFFECT\* NAME = F2 , LINECNT = 55  
 \*STATISTICS\* SOURCE STATEMENTS = 8, PROGRAM SIZE = 520  
 \*STATISTICS\* NO DIAGNOSTICS GENERATED

FORTRAN IV G LEVEL 21

F3 DATE = 77166  
PAGE 0001  
13/51/23

```
FUNCTION F3(X,XD,P1,P2,P3,DELA,DELD,DELDD)
REAL L1,L2,L3,L4,MR,NU,I,J
COMMON/RLK2/DI,F,HI,I,J,R,S,TI,U
COMMON/RLK4/A,R,C,D,G,CD,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA
COMMON/RLK5/A1,A2,A3,A4,A5,R1,R2,R3,R4,R5
RULK=(N*P2+132500.)  
IF(X.GE.1.) RS1.  
IF(X.LT.-1.) RS2;  
F3=RULK*(1.*X+2.*X+3.*1416*L4*((A+DELA)*DELA*(D+DELD)*DELDD)+R*A5*  
1XD)+U*(A4*XD-QF(P1,P2,X))-J*(A4*XD*QR(P2,P3,X))/((3.1416*L4*  
2*((A+DELA)**2-(D+DELD)**2)+V0*A5*(X-(L1+L2+L3))-J*(A3*AR(X))*X*  
3U*(A4*AF(X)*(X-(L1+L2+L3)))  
RETURN  
END
```

0001  
0002  
0003  
0004  
0005  
0006  
0007  
0008  
0009  
0010  
0011

FORTRAN IV G LEVEL 21

F3 DATE = 77166  
PAGE 0002  
13/51/23

```
*OPTIONS IN EFFECT* NOID, EBCDIC, SOURCE, NOLIST, NODECK, LOAD, NOMAP
*OPTIONS IN EFFECT* NAME = F3 , LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 11, PROGRAM SIZE = 970
*STATISTICS* NO DIAGNOSTICS GENERATED
```

FORTRAN IV G LEVEL 21

DATE = 77166 13/51/23 PAGE 0001

```
0001      FUNCTION F4(X,XD,P2,P3)
0002        REAL L1,L2,L3,L4,MR,N,NU,I,J
0003        COMMON/RLK2/DI,F,HI,I,J,R,S,TI,U
0004        COMMON/BLK4/A,B,C,D,G,CD,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA
0005        COMMON/BLK5/A1,A2,A3,A4,A5,R1,R2,R3,R4,R5
0006        BULK=N*P3*132500.
0007        F4=-J*BULK*(A3*XD+QR(P2,P3,X))/((A3+AR(X))*X)
0008        RETURN
0009      END
```

FORTRAN IV G LEVEL 21

DATE = 77166 13/51/23 PAGE 0002

```
*OPTIONS IN EFFECT* NOID,EBCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME = F4   , LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 9,PROGRAM SIZE = 514
*STATISTICS* NO DIAGNOSTICS GENERATED
```

```

      FORTRAN IV G LEVEL 21          DATE = 77166          13/51/23
      0001      FUNCTION F5(DELA,DELAD,P2)
      0002      REAL L1,L2,L3,L4,MN,NU,I,J
      0003      COMMON/RLK4/A,R,C,D,G,CD,RHOS,L1,L2,L3,L4,E,N,NU,VN,MR,GAMMA
      0004      F5=1.0*(RHOS*(R**2-1.0))+(2.0*A**2-2.0)*SINFLA/(R**2
      1+(1.0*NU)*A**2)/(B**2+A**2+NU))
      0005      RETURN
      0006      END

```

PAGE 0002  
13/51/23  
FORTRAN IV G LEVEL 21  
FS DATE = 77166  
\*OPTIONS IN EFFECT\* NOID, EBCDIC, SOURCE, NOLIST, NODECK, LOAD, NOMAP  
\*OPTIONS IN EFFECT\* NAME = FS , LINECNT = 55  
\*STATISTICS\* SOURCE STATEMENTS = 6, PROGRAM SIZE = 472  
\*STATISTICS\* NO DIAGNOSTICS GENERATED

FORTRAN IV G LEVEL 21 F6 DATE = 77166 PAGE 0001  
 13/51/23  
 0001 FUNCTION F6(DELD,DELD0,P2)  
 0002 REAL L1,L2,L3,L4,MR,N,NU,I,J  
 0003 COMMON/RLK4/A,R,C,D,G,CD,RHOF,RHOS,L1,L2,L3,L4,E,N,NU,V0,MR,GAMMA  
 0004 F6=1. / (RHOS\*(D\*\*2-C\*\*2)) \* (-2.\*D\*\*P2-2.\*E\*DELD/(D\*\*2\*(1.-NU)\*  
 1C\*\*2)/(D\*\*2-C\*\*2)-NU))  
 0005 RETURN  
 0006 END

FORTRAN IV G LEVEL 21 F6 DATE = 77166 PAGE 0002  
 13/51/23  
 \*OPTIONS IN EFFECT\* NOID,EBCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP  
 \*OPTIONS IN EFFECT\* NAME = F6 LINECNT = 55  
 \*STATISTICS\* SOURCE STATEMENTS = 6,PROGRAM SIZE = 472  
 \*STATISTICS\* NO DIAGNOSTICS GENERATED

FORTRAN IV G LEVEL 21

```
0001      FUNCTION F7(A*DELA,DELAD,P2)
0002      F7=(A*DELA)*DELAD*P2
0003      RETURN
0004      END
```

PAGE 0001  
13/51/23

DATE = 77166

F7

FORTRAN IV G LEVEL 21

```
*OPTIONS IN EFFECT* NOID,ERCODE,SOURCE,NOLIST,NOECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME=F7
*STATISTICS* SOURCE STATEMENTS= 55
*STATISTICS* NO DIAGNOSTICS GENERATED 4,PROGRAM SIZE= 374
```

PAGE 0002  
13/51/23

DATE = 77166

F7

FORTRAN IV G LEVEL 21

DATE = 77166

13/51/23

PAGE 0001

```
0001      FUNCTION F8(D•DELD•DELDD•P2)
0002      F8=(D•DELD)•DELDD•P2
0003      RETURN
0004      END
```

FORTRAN IV G LEVEL 21

DATE = 77166

13/51/23

PAGE 0002

```
*OPTIONS IN EFFECT* NOID•EBCDIC•SOURCE•NOLIST•NODECK•LOAD•NOMAP
*OPTIONS IN EFFECT* NAME = F8   , LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 4, PROGRAM SIZE = 374
*STATISTICS* NO DIAGNOSTICS GENERATED
```

FORTRAN IV G LEVEL 21

```
0001      F9
0002      FUNCTION F9(V,P2,P2D,N)
0003      REAL L1,L2,L3,L4,MR,N,NU,I,J
0004      F9=V*P2*P2D/(N*N*P2*(I+J))
0005      RETURN
0006      END
```

DATE = 77166

PAGE 0001  
13/51/66

FORTRAN IV G LEVEL 21

F9

DATE = 77166

PAGE 0002

```
*OPTIONS IN EFFECT* NOID,ERCDIC,SOURCE+NOLIST+NODECK+LOAD,NOMAP
*OPTIONS IN EFFECT* NAME=F9
*STATISTICS* SOURCE STATEMENTS = 55
*STATISTICS* PROGRAM SIZE = 388
*STATISTICS* NO DIAGNOSTICS GENERATED
```

FORTRAN IV G LEVEL 21 LINEAR DATE = 77166 13/51/23  
 PAGE 0001  
 0001 SURROUNGE LINEAR(A,X,Y,VV)  
 0002 DIMENSION X(81),Y(81)  
 0003 I=1  
 C 1 IF(Y(I+1) .LT. Y(I)) GO TO 100  
 C USE FOLLOWING IF AS Y INCREASES X INCREASES  
 0004 10 IF(A-X(I))3,2,2  
 C USE FOLLOWING IF AS Y INCREASES X DECREASES  
 C 100 IF(A-X(I))2,2,3  
 0005 2 I=I+1  
 0006 60 T0 10  
 0007 3 I=I-1  
 0008 VV=Y(I)\*(A-X(I+1))/(X(I)-X(I+1))+Y(I+1)\*(A-X(I))/(X(I+1)-X(I))  
 RETURN  
 0009 END  
 0010

FORTRAN IV G LEVEL 21 LINEAR DATE = 77166 13/51/23  
 PAGE 0002  
 \*OPTIONS IN EFFECT\* NOID,EBCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP  
 \*OPTIONS IN EFFECT\* NAME = LINEAR , LINECNT = 55  
 \*STATISTICS\* SOURCE STATEMENTS = 10,PROGRAM SIZE = 506  
 \*STATISTICS\* NO DIAGNOSTICS GENERATED

FORTRAN IV G LEVEL 21 SGN DATE = 77166  
PAGE 0001

0001 FUNCTION SGN(XX)  
0002 IF(XX) 1,2,2  
0003 1 SGN=-1.  
0004 60 TO 3  
0005 2 SGN=1.  
0006 3 RETURN  
0007 END

FORTRAN IV G LEVEL 21 SGN DATE = 77166  
PAGE 0002

\*OPTIONS IN EFFECT\* NOID,ERCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP  
\*OPTIONS IN EFFECT\* NAME = SGN , LINECNT = 55  
\*STATISTICS\* SOURCE STATEMENTS = 7,PROGRAM SIZE = 346  
\*STATISTICS\* NO DIAGNOSTICS GENERATED

FORTRAN IV G LEVEL 21

DATE = 77166

13/51/23

PAGE 0001

```
0001      SUBROUTINE KUTTA(X,XD,P1,P2)
0002      COMMON/RLK1/H,HH
0003      AK1=H*F1(X,XD,P1,P2)
0004      ZX=X+HH*XD+H*AK1/8.
0005      ZX0=XD+AK1/2
0006      AK2=H*F1(ZX,ZX0,P1,P2)
0007      ZX0=XD+AK2/2.
0008      AK3=H*F1(ZX,ZX0,P1,P2)
0009      ZX=X+H*XD+HH*AK3
0010      ZX0=XD+AK3
0011      AK4=H*F1(ZX,ZX0,P1,P2)
0012      X=X+H*(XD+(AK1+AK2+AK3)/6.)
0013      XD=XD+(AK1+2.* (AK2+AK3)+AK4)/6.
0014      RETURN
0015      END
```

KUTTA

FORTRAN IV G LEVEL 21

DATE = 77166

13/51/23

PAGE 0002

```
*OPTIONS IN EFFECT* NOID,ERCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME = KUTTA , LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 15,PROGRAM SIZE = 690
*STATISTICS* NO DIAGNOSTICS GENERATED
```

**FORTRAN IV G LEVEL 21**

DATE = 77166

PAGE 0001

```
0001      FUNCTION PRF(XXX)
0002      COMMON/ALK3/IP,PB1,PB2,PB3
0003      IF(IP.GT.1) GO TO 1
0004      PRF=PB1
0005      IP=IP+1
0006      RETURN
0007      1 IF(IP.GT.2) GO TO 2
0008      PBF=PB2
0009      IP=IP+1
0010      RETURN
0011      2 IF(IP.GT.3) GO TO 3
0012      PRF=PB2
0013      IP=IP+1
0014      RETURN
0015      3 PBF=PB3
0016      4 RETURN
0017      END
```

13/51/23

**FORTRAN IV G LEVEL 21**

DATE = 77166

PAGE 0002

```
*OPTIONS IN EFFECT*  NOID,ERCDIC,SOURCE,NOLIST,NODECK,LOAD,NOMAP
*OPTIONS IN EFFECT*  NAME = PBF   LINECNT = 55
*STATISTICS*  SOURCE STATEMENTS = 17,PROGRAM SIZE = 482
*STATISTICS*  NO. DIAGNOSTICS GENERATED
```

13/51/23

FORTRAN IV G LEVEL 21 DQ64 DATE = 77166 PAGE 0001

```
0001      SURROUNTE DQ64 (XL,XU,FCT,Y)
0002      A=.5*(XU+XL)
0003      B=XU-XL
0004      C=.43056815579702629*B
0005      Y=.17392742256872693*(FCT(A+C)+FCT(A-C))
0006      C=.16999052179242813*B
0007      Y=B*(Y+.32607257743127307*(FCT(A+C)+FCT(A-C)))
0008      RETURN
0009      END
```

13/51/23

FORTRAN IV G LEVEL 21 DQ64 DATE = 77166 PAGE 0002

```
*OPTIONS IN EFFECT* NOTD,ERCDIC,SOURCE,NOLIST,NODECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME = DQ64 , LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 9,PROGRAM SIZE = 586
*STATISTICS* NO DIAGNOSTICS GENERATED
```

13/51/23

FORTRAN IV G LEVEL 21

```
      KUTTA1          DATE = 77166
      13/51/23
0001   SUBROUTINE KUTTA1(X,XD,P1,P2,P3,DELA,DELAD,DELD,DELD)
0002     COMMON/ALK1/H,HH
0003     AK1=H*F2(X,XD,P1,P2)
0004     AL1=H*F3(X,XD,P1,P2,P3,DELA,DELAD,DELD,DELD)
0005     AM1=H*F4(X,XD,P2,P3)
0006     ZP1=P1+AK1/2.
0007     ZP2=P2+AL1/2.
0008     ZP3=P3+AM1/2.
0009     AK2=H*F2(X,XD,ZP1,ZP2)
0010     AL2=H*F3(X,XD,ZP1,ZP2,ZP3,DELA,DELAD,DELD,DELD)
0011     AM2=H*F4(X,XD,ZP2,ZP3)
0012     ZP1=P1+AK2/2.
0013     ZP2=P2+AL2/2.
0014     ZP3=P3+AM2/2.
0015     AK3=H*F2(X,XD,ZP1,ZP2)
0016     AL3=H*F3(X,XD,ZP1,ZP2,ZP3,DELA,DELAD,DELD,DELD)
0017     AM3=H*F4(X,XD,ZP2,ZP3)
0018     ZP1=P1+AK3
0019     ZP2=P2+AL3
0020     ZP3=P3+AM3
0021     AK4=H*F2(X,XD,ZP1,ZP2)
0022     AL4=H*F3(X,XD,ZP1,ZP2,ZP3,DELA,DELAD,DELD,DELD)
0023     AM4=H*F4(X,XD,ZP2,ZP3)
0024     P1=P1+1./6.* (AK1+2.*AK2+2.*AK3+AK4)
0025     P2=P2+1./6.* (AL1+2.*AL2+2.*AL3+AL4)
0026     P3=P3+1./6.* (AM1+2.*AM2+2.*AM3+AM4)
0027     RETURN
0028 END
```

FORTRAN IV G LEVEL 21

KUTTA1 DATE = 77166

13/51/23

```
*OPTIONS IN EFFECT* NOID,ERCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME = KUTTA1    * LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 28,PROGRAM SIZE = 1308
*STATISTICS* NO DIAGNOSTICS GENERATED
```

PAGE 0001

PAGE 0002

PAGE 0003

FORTRAN IV G LEVEL 21

KUTTA2 DATE = 77166

13/51/23

PAGE 0001

```
0001      SURROUNIQUE KUTTA2 (DELA•DELD•DELAD•DELDD•P2)
          COMMON/RLK1/H•HH
          AK1=H•F5 (DELA•DELAD•P2)
          AL1=H•F6 (DELD•DELD•P2)
          ZDELA=DELA•HH•DELAD•H•AK1/8.
          ZDELD=DELD•HH•DELD•H•AL1/8.
          ZDELAD=DELAD•AK1/2.
          ZDELDD=DELDD•AL1/2.
          AK2=H•F5 (ZDELA•ZDELAD•P2)
          AL2=H•F6 (ZDELD•ZDELDD•P2)
          ZDELAD=DELAD•AK2/2.
          ZDELDD=DELDD•AL2/2.
          AK3=H•F5 (ZDELA•ZDELAD•P2)
          AL3=H•F6 (ZDELD•ZDELDD•P2)
          ZDELA=DELA•H•DELAD•HH•AK3
          ZDELDD=DELDD•H•DELDD•HH•AL3
          ZDELAD=DELAD•AK3
          ZDELDD=DELDD•AL3
          AK4=H•F5 (ZDELA•ZDELAD•P2)
          AL4=H•F6 (ZDELD•ZDELDD•P2)
          DELA=DELA•H•(DELAD•(AK1•AK2•AK3)/6.)
          DELD=DELD•H•(DELDD•(AL1•AL2•AL3)/6.)
          DELAD=DELAD•(AK1•2•(AK2•AK3•AK4)/6.
          DELDD=DELDD•(AL1•2•(AL2•AL3•AL4)/6.
          RETURN
END-
```

FORTRAN IV G LEVEL 21

KUTTA2

13/51/23

DATE = 77166

PAGE 0002

```
*OPTIONS IN EFFECT* NOID,ERCDIC,SOURCE,NOLIST,NOECK,LOAD,NOMAP
*OPTIONS IN EFFECT* NAME = KUTTA2   • LINECNT = 55
*STATISTICS* SOURCE STATEMENTS = 26,PROGRAM SIZE = 1012
*STATISTICS* NO DIAGNOSTICS GENERATED
*STATISTICS* NO DIAGNOSTICS THIS STEP
```

F88-LEVEL LINKAGE EDITOR OPTIONS SPECIFIED LIST.LET.XREF  
 \*\*\*\*\*MAIN DEFAULT OPTION(S) USED - SIZE=(92160,8192)  
 \*\*\*\*\*DOES NOT EXIST BUT HAS BEEN ADDED TO DATA SET

### CROSS REFERENCE TABLE

CONTROL SECTION				ENTRY				LOCATION			
NAME	ORIGIN	LENGTH		NAME	LOCATION	NAME	LOCATION	NAME	LOCATION	NAME	LOCATION
MAIN	00	157C									
AF	1580	1E4									
AR	1768	17C									
QF	18E8	10E									
QR	1AC8	10E									
F1	1CA8	26A									
F2	1FB8	208									
F3	2129	3CA									
F4	24F0	272									
F5	26F8	17A									
F6	28D0	1nA									
F7	2AA8	176									
F8	2C20	176									
F9	2D98	1A4									
LINIFAR	2F20	1FA									
5GN	3126	15A									
KUTTA	3280	2B2									
PBF	3538	1E2									
DG64	3720	24A									
KUTTA1	3970	51C									
KUTTA2	3E90	3F4									
IHMEXIT*	4288	1C									
				EXIT	4288						
IHMNAMEL*	42A8	A4F									
IHMCOMM*	4D58	F8C		FRDNL#	42A8	FWRNL#	4898				
IHMCOMM2*	5CE8	715		IBCOM#	40R4	FDI0CS#	4E40	INTSWITCH	5C8E		
IHMSSCN*	6400	208		SEQDASD	6116						
IHMSSQRT*	6608	168		COS	6400	SIN	6422				
IHMFCVTH*	6770	A6F		SQRT	6608						
				ADCON#	6770	FCVAOUTP	681A				
				FCVIOUTP	6DAE	FCVEDOUTP	6FEA				
IHMFCNTH*	72E0	548		ARITH#	72E0	ADJSWITCH	767C				
IHMFIOS*	7828	FF8		FI0CS#	7828	FI0CSBEP	782E				
IHMFIOS2*	8820	5B8		ERRMON	8008	IHMERR	8DF0				
IHMERRM*	8008	5FC									

	NAME	ORIGIN	LENGTH		NAME	LOCATION		NAME	LOCATION		NAME	LOCATION
	IHNNUOPT *	93D8	318		FQCONI#	96F0						
	IHNFCONI*	96F0	2E5									
	IHNFCONO*	99D8	442		FQCONO#							
	IHNTRCH*	9E80	2A6		IHNTRCH							
	IHNNUATBL*	A128	638									
	IHNFTEN *	A760	198		FTEH#							
	BLK1	A8FB	R									
	BLK2	A900	24									
	BLK3	A928	10									
	BLK4	A938	48									
	BLK5	A980	2A									

	LOCATION	REFERS TO SYMBOL	IN CONTROL SECTION		LOCATION	REFERS TO SYMBOL	IN CONTROL SECTION		LOCATION	REFERS TO SYMBOL	IN CONTROL SECTION
	3A8	BLK4	BLK4		3AC	BLK4	BLK4				
	39C	BLK4	BLK4		3B8	BLK3	BLK3				
	3C8	BLK3	BLK3		3D8	BLK3	BLK3				
	198	F7	F7		19C	F8	F8				
	1A0	F9	F9		1A4	IBCOM#	IBCOM#				
	1A8	FRDNL#	IHNNAMEL		1AC	FWRNL#	FWRNL#				
	1R0	LINEAR	LINEAR		1B4	KUTTA	KUTTA				
	1R8	F1	F1		1BC	KUTTA2	KUTTA2				
	1C0	F5	F5		70C	BLKS	BLKS				
	71C	BLK5	BLK5		72C	BLKS	BLKS				
	73C	BLK5	BLK5		74C	BLK4	BLK4				
	75C	BLK4	BLK4		76C	BLK4	BLK4				
	77C	BLK4	BLK4		78C	BLK4	BLK4				
	79C	BLK4	BLK4		7AC	BLK4	BLK4				
	7BC	BLK4	BLK4		7CC	BLK4	BLK4				
	7DC	BLK4	BLK4		7EC	BLK4	BLK4				
	7FC	BLK4	BLK4		80C	BLK4	BLK4				
	81C	BLK4	BLK4		82C	BLK4	BLK4				
	84C	BLK4	BLK4		78	BLK1	BLK1				
	7C	BLK2	BLK2		80	BLK3	BLK3				
	84	BLK4	BLK4		88	BLK5	BLK5				
	1EC	BLK3	BLK3		1FC	BLK5	BLK5				
	20C	BLK3	BLK3		1C4	F6	F6				
	1C8	KUTTA1	KUTTA1		1CC	F2	F2				
	1D0	F3	F3		1D4	F4	F4				
	1D8	EXIT	IHNFEEXIT		1DC	SQRT	IHNSSQRT				
	6FC	BLK5	BLK5		15FC	BLK5	BLK5				
	15FB	BLK4	BLK4		17E4	BLK5	BLK5				
	17E0	BLK4	BLK4		197C	AF	AF				
	1980	SGN	SGN		1984	SQRT	IHNSSQRT				
	1960	BLK4	BLK4		1B5C	AR	AR				
	1860	SGN	SGN		1B64	IHNSSQRT	IHNSSQRT				

LOCATION	REFERS TO SYMBOL	IN CONTROL SECTION
1840	BLK4	BLK4
1D28	BLK5	BLK4
1D4C	PRF	SIN
1D20	RLK2	BLK4
2570	BLK5	BLK4
2590	AR	QR
2770	RLK4	BLK2
1F94	BLK4	BLK4
1FR4	QF	BLK5
1F90	BLK2	BLK4
21A0	BLK5	BLK4
39F8	GR	QF
21CB	AF	AR
21D0	F1	AF
3314	BLK3	BLK2
35B0	F3	F1
3A08	BLK1	BLK1
42A0	F6	BLK1
4B30	IHC0MH	IBCOMH
4C14	ADCON#	IHFCS#
4E98	IHNERRM	IHNFIOS
5RAC	IHNCOMH2	IHNCOMH2
5BE8	F1OCS#	SEDDASD
5CE0	ADJSWTC	ADCON#
5RD0	IHCMSF	ARITH#
5RD8	FCVLOUTP	SRC8
5RE0	FCVCOUTP	SRE4
589C	FCVZOUTP	SACC
5R74	IHNCOMH2	SB04
5B7C	IHNCOMH2	SD0C
5FB0	IHNCOMH	IHNFIOT
5FD0	IHNCOMH	IHNFIOT
5D2C	IHC0MH	IHNFIOT
626D	IHNCOMH	IHNFIOT
6550	IHC0MH	IHNFIOT
66EC	IHC0MH	IHNFIOT
70D0	IHC0MH	IHNFIOT
7120	FQCONO#	IHNFIOT
76CC	IHC0MH	IHNFIOT
7678	INT6SWCH	IHNFIOT
76D8	ADCON#	IHNFIOT
7744	IHNERRM	IHNFIOT
798C	IHNFIOS2	IHNFIOT
863B	IHC0MH	IHNFIOT
8664	IHNFIOS2	IHNFIOT
93C4	IHNFIOS2	IHNFIOT
93CC	IHNTRCH	IHNFIOT
996C	FTEN#	IHNFTEN
A010	LDF10#	IBCOM#
A00C	ADCON#	IHNFIOS
		FIOSCBE#
		\$UNRESOLVED(W)

LOCATION	REFERS TO SYMBOL	IN CONTROL SECTION
1D24	BLK4	BLK4
1D58	BLK5	BLK4
1D50	PRF	SIN
256C	RLK2	BLK4
258C	BLK5	BLK4
2568	AR	BLK2
2948	BLK4	BLK4
1F98	BLK5	BLK5
1FB8	QF	BLK4
219C	BLK2	BLK4
21C4	GR	QF
21CC	AF	AR
2198	F1	AF
32F8	BLK1	BLK1
3A04	F3	F2
3A0C	BLK1	F4
3F24	F6	F5
3F08	IHC0MH	IBCOMH
4B28	IHFCS#	IHNFIOS
4B2C	IHNERRM	IHNFIOT
4E40	IHNCOMH2	IHNFIOT
5RC4	IHNFIOS	IHNFIOT
5BDC	IHNFIOT	IHNFIOT
5B70	IHNFIOT	IHNFIOT
5B40	IHNFIOT	IHNFIOT
5B78	IHNFIOT	IHNFIOT
5B80	IHNFIOT	IHNFIOT
5FD9	IHNFIOT	IHNFIOT
5D30	IHNFIOT	IHNFIOT
625D	IHNFIOT	IHNFIOT
627D	IHNFIOT	IHNFIOT
6594	IHNFIOT	IHNFIOT
6714	IHNFIOT	IHNFIOT
70CC	IHNFIOT	IHNFIOT
7124	IHNFIOT	IHNFIOT
76D0	IHNFIOT	IHNFIOT
7674	IHNFIOT	IHNFIOT
76D4	IHNFIOT	IHNFIOT
798B	IHNFIOT	IHNFIOT
862C	IHNFIOT	IHNFIOT
864D	IHNFIOT	IHNFIOT
8819	IHNFIOT	IHNFIOT
93C8	IHNFIOT	IHNFIOT
93D0	IHNTRCH	IHNFIOT
9DC8	FTEN#	IHNFTEN
A008	LDF10#	IBCOM#
A014	ADCON#	IHNFIOS

ENTRY ADDRESS  
TOTAL LENGTH

00  
4948



T=0.0339 X=21.168 XD= 69.54 XDD= 1975.4 P1= 2865.8A P1D= 0.0 P2= 2865.8B P2D= -7733.6 P3= 2865.8B P3D= 0.0  
 DELA= 0.0188 DELAD=-0.0184 DELD= 0.0184 DELD0= 0.0184 SIGT= 55205.4 SIGAX= 2616.8 SIGR= 2849.89 SIG= 35561.2  
 T=0.0359 X=21.330 XD= 73.48 XDD= 1962.9 P1= 2849.89 P1D= 0.0 P2= 2849.89 P2D= -8276.5 P3= 2849.89 P3D= 0.0  
 DELA= 0.0187 DELAD=-0.0189 DELD= 0.0189 DELD0= 0.0189 SIGT= 54897.3 SIGAX= 2602.7 SIGR= -2849.9 SIG= 35332.8  
 T=0.0379 X=21.481 XD= 77.39 XDD= 1949.7 P1= 2833.04 P1D= 0.0 P2= 2833.04 P2D= -8524.9 P3= 2833.04 P3D= 0.0  
 DELA= 0.0186 DELAD=-0.0187 DELD= 0.0187 DELD0= 0.0187 SIGT= 54572.7 SIGAX= 25869.9 SIGR= -2833.0 SIG= 35153.7  
 T=0.039 X=21.634 XD= H1.2A XDD= 1935.8 P1= 2815.33 P1D= 0.0 P2= 2815.33 P2D= -9022.3 P3= 2A15.33 P3D= 0.0  
 DELA= 0.0185 DELAD=-0.0186 DELD= 0.0186 DELD0= 0.0186 SIGT= 54231.6 SIGAX= 2508.1 SIGR= -2815.3 SIG= 34933.9  
 T=0.0419 X=21.805 XD= 85.13 XDD= 1921.3 P1= 2796.78 P1D= 0.0 P2= 2796.78 P2D= -9504.4 P3= 2796.78 P3D= 0.0  
 DELA= 0.0184 DELAD=-0.0185 DELD= 0.0185 DELD0= 0.0185 SIGT= 53874.2 SIGAX= 25338.8 SIGR= -2796.8 SIG= 34933.9  
 T=0.0439 X=21.979 XD= AB.96 XDD= 1904.1 P1= 2777.0 P1D= 0.0 P2= 2777.0 P2D= -9923.0 P3= 2777.0 P3D= 0.0  
 DELA= 0.0182 DELAD=-0.0180 DELD= 0.0180 DELD0= 0.0180 SIGT= 53500.9 SIGAX= 2691.7 SIGR= -2777.4 SIG= 34463.3  
 T=0.0459 X=22.161 XD= 92.76 XDD= 1890.3 P1= 2757.19 P1D= 0.0 P2= 2757.19 P2D= -10235.4 P3= 2757.19 P3D= 0.0  
 DELA= 0.0181 DELAD=-0.0184 DELD= 0.0184 DELD0= 0.0184 SIGT= 53111.7 SIGAX= 25177.3 SIGR= -2757.2 SIG= 34212.5  
 T=0.0479 X=22.350 XD= 96.52 XDD= 1873.8 P1= 2736.16 P1D= 0.0 P2= 2736.16 P2D= -10696.6 P3= 2736.16 P3D= 0.0  
 DELA= 0.0180 DELAD=-0.0186 DELD= 0.0186 DELD0= 0.0186 SIGT= 52706.6 SIGAX= 24985.2 SIGR= -2736.2 SIG= 34951.6  
 T=0.0499 X=22.546 XD= 100.25 XDD= 1856.7 P1= 2714.34 P1D= 0.0 P2= 2714.34 P2D= -11183.4 P3= 2714.34 P3D= 0.0  
 DELA= 0.0178 DELAD=-0.0165 DELD= 0.0165 DELD0= 0.0165 SIGT= 52286.2 SIGAX= 24786.0 SIGR= -2714.3 SIG= 34680.8  
 T=0.0519 X=22.750 XD= 103.95 XDD= 1839.0 P1= 2691.72 P1D= 0.0 P2= 2691.72 P2D= -11465.9 P3= 2691.72 P3D= 0.0  
 DELA= 0.0177 DELAD=-0.0164 DELD= 0.0164 DELD0= 0.0164 SIGT= 51850.6 SIGAX= 24579.4 SIGR= -2691.7 SIG= 34212.5  
 T=0.0539 X=22.961 XD= 101.60 XDD= 1820.6 P1= 2668.32 P1D= 0.0 P2= 2668.32 P2D= -11841.9 P3= 2668.32 P3D= 0.0  
 DELA= 0.0175 DELAD=-0.0171 DELD= 0.0171 DELD0= 0.0171 SIGT= 51399.7 SIGAX= 24365.7 SIGR= -2668.3 SIG= 33109.7  
 T=0.0559 X=23.180 XD= 111.23 XDD= 1801.7 P1= 2644.14 P1D= 0.0 P2= 2644.14 P2D= -12340.7 P3= 2644.14 P3D= 0.0  
 DELA= 0.0174 DELAD=-0.0173 DELD= 0.0173 DELD0= 0.0173 SIGT= 50934.1 SIGAX= 24145.0 SIGR= -2644.1 SIG= 33109.7  
 T=0.0579 X=23.405 XD= 114.81 XDD= 1762.1 P1= 2619.22 P1D= 0.0 P2= 2619.22 P2D= -12661.7 P3= 2619.22 P3D= 0.0  
 DELA= 0.0172 DELAD=-0.0172 DELD= 0.0172 DELD0= 0.0172 SIGT= 50454.0 SIGAX= 23914.6 SIGR= -2691.7 SIG= 33109.8  
 T=0.0599 X=23.638 XD= 118.35 XDD= 1762.0 P1= 2593.54 P1D= 0.0 P2= 2593.54 P2D= -12956.3 P3= 2593.54 P3D= 0.0  
 DELA= 0.0170 DELAD=-0.0181 DELD= 0.0181 DELD0= 0.0181 SIGT= 49959.3 SIGAX= 23682.9 SIGR= -2568.3 SIG= 33109.7  
 T=0.0619 X=23.878 XD= 121.85 XDD= 1741.3 P1= 2567.13 P1D= 0.0 P2= 2567.13 P2D= -13380.2 P3= 2567.13 P3D= 0.0  
 DELA= 0.0169 DELAD=-0.0182 DELD= 0.0182 DELD0= 0.0182 SIGT= 49450.5 SIGAX= 23441.7 SIGR= -2567.1 SIG= 33109.8  
 T=0.0639 X=24.125 XD= 125.32 XDD= 1720.0 P1= 2540.00 P1D= 0.0 P2= 2540.00 P2D= -13757.2 P3= 2540.00 P3D= 0.0  
 DELA= 0.0167 DELAD=-0.0183 DELD= 0.0183 DELD0= 0.0183 SIGT= 48927.9 SIGAX= 23194.0 SIGR= -2564.0 SIG= 32500.5  
 T=0.0659 X=24.379 XD= 128.73 XDD= 1698.2 P1= 2512.16 P1D= 0.0 P2= 2512.16 P2D= -14044.1 P3= 2512.16 P3D= 0.0  
 DELA= 0.0165 DELAD=-0.0187 DELD= 0.0187 DELD0= 0.0187 SIGT= 49959.3 SIGAX= 22939.8 SIGR= -2512.2 SIG= 32181.9  
 T=0.0679 X=24.639 XD= 132.11 XD= 1675.8 P1= 2483.62 P1D= 0.0 P2= 2483.62 P2D= -14453.8 P3= 2483.62 P3D= 0.0  
 DELA= 0.0163 DELAD=-0.0188 DELD= 0.0188 DELD0= 0.0188 SIGT= 47842.0 SIGAX= 22619.2 SIGR= -2483.6 SIG= 32181.9  
 T=0.0699 X=24.907 XD= 135.43 XDD= 1652.9 P1= 2454.41 P1D= 0.0 P2= 2454.41 P2D= -15044.4 P3= 2454.41 P3D= 0.0  
 DELA= 0.0161 DELAD=-0.0186 DELD= 0.0186 DELD0= 0.0186 SIGT= 48927.9 SIGAX= 22442.4 SIGR= -2454.4 SIG= 31511.5  
 T=0.0719 X=25.180 XD= 138.72 XDD= 1629.5 P1= 2424.53 P1D= 0.0 P2= 2424.53 P2D= -15072.2 P3= 2424.53 P3D= 0.0  
 DELA= 0.0159 DELAD=-0.0193 DELD= 0.0193 DELD0= 0.0193 SIGT= 48339.17 SIGAX= 2239.6 SIGR= -2512.2 SIG= 31172.1  
 T=0.0739 X=25.461 XD= 141.95 XDD= 1605.6 P1= 2483.62 P1D= 0.0 P2= 2483.62 P2D= -14453.8 P3= 2483.62 P3D= 0.0  
 DELA= 0.0157 DELAD=-0.0187 DELD= 0.0187 DELD0= 0.0187 SIGT= 46703.6 SIGAX= 22939.99 SIGR= -2483.6 SIG= 31172.1  
 T=0.0759 X=25.748 XD= 145.14 XDD= 1581.1 P1= 2454.41 P1D= 0.0 P2= 2454.41 P2D= -15044.4 P3= 2454.41 P3D= 0.0  
 DELA= 0.0155 DELAD=-0.01958 DELD= 0.01958 DELD0= 0.01958 SIGT= 47279.2 SIGAX= 22442.4 SIGR= -2454.4 SIG= 31511.5  
 T=0.0779 X=26.041 XD= 148.27 XDD= 1556.2 P1= 2331.02 P1D= 0.0 P2= 2331.02 P2D= -15072.2 P3= 2331.02 P3D= 0.0  
 DELA= 0.0153 DELAD=-0.01991 DELD= 0.01991 DELD0= 0.01991 SIGT= 46703.6 SIGAX= 2239.6 SIGR= -2454.4 SIG= 30984.7  
 T=0.0799 X=26.340 XD= 151.36 XDD= 1530.8 P1= 2299.61 P1D= 0.0 P2= 2299.61 P2D= -15614.6 P3= 2299.61 P3D= 0.0  
 DELA= 0.0151 DELAD=-0.0107 DELD= 0.0107 DELD0= 0.0107 SIGT= 461115.4 SIGAX= 21860.7 SIGR= -2483.6 SIG= 30984.7  
 T=0.0809 X=26.646 XD= 154.39 XDD= 151.39 P1= 2352.82 P1D= 0.0 P2= 2352.82 P2D= -15772.2 P3= 2352.82 P3D= 0.0  
 DELA= 0.0149 DELAD=-0.01044 DELD= 0.01044 DELD0= 0.01044 SIGT= 45514.9 SIGAX= 21576.0 SIGR= -2454.4 SIG= 30984.7  
 T=0.0829 X=26.957 XD= 157.38 XDD= 1478.6 P1= 2222.02 P1D= 0.0 P2= 2222.02 P2D= -16011.8 P3= 22331.02 P3D= 0.0  
 DELA= 0.0147 DELAD=-0.01043 DELD= 0.01043 DELD0= 0.01043 SIGT= 44902.3 SIGAX= 21285.7 SIGR= -2331.0 SIGE= 28924.4  
 T=0.0859 X=27.274 XD= 160.31 XDD= 1451.8 P1= 2197.87 P1D= 0.0 P2= 2197.87 P2D= -17198.1 P3= 2197.87 P3D= 0.0  
 DELA= 0.0144 DELAD=-0.0157 DELD= 0.0157 DELD0= 0.0157 SIGT= 44237.0 SIGAX= 20989.7 SIGR= -2298.6 SIGE= 2905.8  
 T=0.0879 X=27.598 XD= 163.18 XDD= 1424.6 P1= 2235.60 P1D= 0.0 P2= 2235.60 P2D= -16675.6 P3= 2265.60 P3D= 0.0  
 DELA= 0.0142 DELAD=-0.01070 DELD= 0.01070 DELD0= 0.01070 SIGT= 43664.2 SIGAX= 20688.3 SIGR= -2262.8 SIG= 2919.0  
 T=0.0899 X=27.926 XD= 166.00 XDD= 1397.0 P1= 2127.95 P1D= 0.0 P2= 2127.95 P2D= -16907.9 P3= 2232.02 P3D= 0.0  
 DELA= 0.0140 DELAD=-0.1088 DELD= 0.01088 DELD0= 0.01088 SIGT= 42995.3 SIGAX= 20381.7 SIGR= -2232.0 SIGE= 27969.0  
 T=0.0919 X=28.261 XD= 168.77 XDD= 1368.9 P1= 2092.20 P1D= 0.0 P2= 2092.20 P2D= -17198.0 P3= 2197.87 P3D= 0.0  
 DELA= 0.0137 DELAD=-0.1057 DELD= 0.01057 DELD0= 0.01057 SIGT= 42337.5 SIGAX= 20069.8 SIGR= -2298.6 SIGE= 27272.2  
 T=0.0939 X=28.601 XD= 171.47 XDD= 1340.5 P1= 2055.95 P1D= 0.0 P2= 2055.95 P2D= -18559.8 P3= 2055.95 P3D= 0.0  
 DELA= 0.0135 DELAD=-0.1115 DELD= 0.01115 DELD0= 0.01115 SIGT= 39603.7 SIGAX= 1873.9 SIGR= -2055.9 SIGE= 25551.2  
 T=0.0959 X=28.946 XD= 174.13 XDD= 1311.7 P1= 2019.22 P1D= 0.0 P2= 2019.22 P2D= -1868.6 P3= 2019.22 P3D= 0.0  
 DELA= 0.0133 DELAD=-0.1137 DELD= 0.01137 DELD0= 0.01137 SIGT= 39961.1 SIGAX= 1843.8 SIGR= -2019.2 SIGE= 25055.4  
 T=0.0979 X=29.297 XD= 176.72 XDD= 1282.5 P1= 1982.02 P1D= 0.0 P2= 1982.02 P2D= -18701.0 P3= 1982.02 P3D= 0.0

DELA= 0.0130 DELAD=-0.1151 DELD= 0.0037 DELDD= 0.0292 SIGT= 38179.5 SIGX= 18098.8 SIGR= -1982.0 SIGE= 24593.8  
 T=0.099 X=29.652 XD= 179.25 XD= 1253.0 P1= 1944.36 PID= 0.0 P2= 1944.36 P2D= -18957.8 P3= 1944.36 P3D= 0.0  
 DELA= 0.0128 DELAD=-0.1157 DELD= 0.0036 DELDD= 0.0296 SIGT= 37454.2 SIGX= 17754.9 SIGR= -1944.4 SIGE= 24156.8  
 T=0.1019 X=30.013 XD= 181.73 XD= 1223.1 P1= 1946.27 PID= 0.0 P2= 1906.27 P2D= -19123.1 P3= 1906.27 P3D= 0.0  
 DELA= 0.0125 DELAD=-0.1180 DELD= 0.0035 DELDD= 0.0300 SIGT= 36720.4 SIGX= 17407.1 SIGR= -1946.3 SIGE= 23653.9  
 T=0.1039 X=30.379 XD= 184.14 XD= 1192.9 P1= 1867.76 P1D= 0.0 P2= 1867.76 P2D= -19348.8 P3= 1867.76 P3D= 0.0  
 DELA= 0.0123 DELAD=-0.1190 DELD= 0.0035 DELDD= 0.0301 SIGT= 35978.5 SIGX= 17055.4 SIGR= -1867.8 SIGE= 23160.0  
 T=0.1059 X=30.749 XD= 186.50 XD= 162.4 P1= 1628.94 P1D= 0.0 P2= 1828.84 P2D= -19586.8 P3= 1828.84 P3D= 0.0  
 DELA= 0.0120 DELAD=-0.1192 DELD= 0.0034 DELDD= 0.0304 SIGT= 35228.9 SIGX= 16700.8 SIGR= -1828.8 SIGE= 22653.1  
 T=0.1079 X=31.124 XD= 188.79 XD= 1131.6 P1= 1789.54 P1D= 0.0 P2= 1789.54 P2D= -19712.9 P3= 1789.54 P3D= 0.0  
 DELA= 0.0118 DELAD=-0.1214 DELD= 0.0033 DELDD= 0.0304 SIGT= 34471.8 SIGX= 16341.1 SIGR= -1789.5 SIGE= 22205.4  
 T=0.1099 X=31.504 XD= 191.02 XD= 1100.5 P1= 1749.86 P1D= 0.0 P2= 1749.86 P2D= -19911.2 P3= 1749.86 P3D= 0.0  
 DELA= 0.0115 DELAD=-0.1225 DELD= 0.0033 DELDD= 0.0311 SIGT= 33707.5 SIGX= 15978.8 SIGR= -1749.9 SIGE= 2173.1  
 T=0.1119 X=31.987 XD= 193.19 XD= 1069.1 P1= 1709.83 P1D= 0.0 P2= 1709.83 P2D= -20130.1 P3= 1709.83 P3D= 0.0  
 DELA= 0.0112 DELAD=-0.1226 DELD= 0.0032 DELDD= 0.0315 SIGT= 32936.4 SIGX= 15613.3 SIGR= -1828.8 SIGE= 21216.4  
 T=0.1139 X=32.276 XD= 195.30 XD= 1037.4 P1= 1669.46 P1D= 0.0 P2= 1669.46 P2D= -20228.1 P3= 1669.46 P3D= 0.0  
 DELA= 0.0110 DELAD=-0.1251 DELD= 0.0031 DELDD= 0.0315 SIGT= 32158.8 SIGX= 15244.7 SIGR= -1669.5 SIGE= 20715.5  
 T=0.1159 X=32.668 XD= 197.34 XD= 1005.5 P1= 1628.77 P1D= 0.0 P2= 1628.77 P2D= -20426.4 P3= 1628.77 P3D= 0.0  
 DELA= 0.0107 DELAD=-0.1251 DELD= 0.0030 DELDD= 0.0320 SIGT= 31375.0 SIGX= 14873.1 SIGR= -1749.9 SIGE= 20210.6  
 T=0.1179 X=33.064 XD= 199.32 XD= 973.4 P1= 1587.78 P1D= 0.0 P2= 1587.78 P2D= -20581.6 P3= 1587.78 P3D= 0.0  
 DELA= 0.0104 DELAD=-0.1258 DELD= 0.0030 DELDD= 0.0322 SIGT= 30585.3 SIGX= 14499.8 SIGR= -1587.8 SIGE= 19701.9  
 T=0.1199 X=33.465 XD= 201.23 XD= 941.0 P1= 1546.49 P1D= 0.0 P2= 1546.49 P2D= -20687.9 P3= 1546.49 P3D= 0.0  
 DELA= 0.0102 DELAD=-0.1275 DELD= 0.01251 DELDD= 0.0323 SIGT= 29790.1 SIGX= 14212.8 SIGR= -1546.5 SIGE= 19189.6  
 T=0.1219 X=33.869 XD= 203.08 XD= 908.4 P1= 1504.94 P1D= 0.0 P2= 1504.94 P2D= -20833.5 P3= 1504.94 P3D= 0.0  
 DELA= 0.0099 DELAD=-0.1276 DELD= 0.0026 DELDD= 0.0325 SIGT= 28989.6 SIGX= 13742.3 SIGR= -1504.9 SIGE= 18674.0  
 T=0.1239 X=34.276 XD= 204.86 XD= 875.6 P1= 1463.13 P1D= 0.0 P2= 1463.13 P2D= -20983.3 P3= 1463.13 P3D= 0.0  
 DELA= 0.0096 DELAD=-0.1290 DELD= 0.0027 DELDD= 0.0322 SIGT= 28184.2 SIGX= 13360.6 SIGR= -1463.1 SIGE= 18155.2  
 T=0.1259 X=34.687 XD= 206.58 XD= 842.6 P1= 1421.08 P1D= 0.0 P2= 1421.08 P2D= -21064.5 P3= 1421.08 P3D= 0.0  
 DELA= 0.0093 DELAD=-0.1297 DELD= 0.0026 DELDD= 0.0329 SIGT= 27374.2 SIGX= 12976.5 SIGR= -1421.1 SIGE= 17633.4  
 T=0.1279 X=35.102 XD= 208.23 XD= 809.5 P1= 1378.80 P1D= 0.0 P2= 1378.80 P2D= -21214.8 P3= 1378.80 P3D= 0.0  
 DELA= 0.0091 DELAD=-0.1294 DELD= 0.0026 DELDD= 0.0332 SIGT= 26559.9 SIGX= 12590.5 SIGR= -1378.8 SIGE= 17108.9  
 T=0.1299 X=35.520 XD= 209.81 XD= 776.1 P1= 1336.33 P1D= 0.0 P2= 1336.33 P2D= -21246.6 P3= 1336.33 P3D= 0.0  
 DELA= 0.0088 DELAD=-0.1315 DELD= 0.0025 DELDD= 0.0332 SIGT= 25741.6 SIGX= 12202.6 SIGR= -1336.3 SIGE= 16581.8  
 T=0.1319 X=35.941 XD= 211.33 XD= 742.7 P1= 1293.65 P1D= 0.0 P2= 1293.65 P2D= -21384.3 P3= 1293.65 P3D= 0.0  
 DELA= 0.0085 DELAD=-0.1310 DELD= 0.0024 DELDD= 0.0334 SIGT= 24919.6 SIGX= 11813.0 SIGR= -1293.7 SIGE= 16052.3  
 T=0.1339 X=36.364 XD= 212.78 XD= 709.1 P1= 1250.81 P1D= 0.0 P2= 1250.81 P2D= -21459.0 P3= 1250.81 P3D= 0.0  
 DELA= 0.0082 DELAD=-0.1318 DELD= 0.0023 DELDD= 0.0335 SIGT= 24094.3 SIGX= 11421.8 SIGR= -1250.8 SIGE= 15520.7  
 T=0.1359 X=36.791 XD= 214.16 XD= 675.3 P1= 1207.81 P1D= 0.0 P2= 1207.81 P2D= -21517.4 P3= 1207.81 P3D= 0.0  
 DELA= 0.0079 DELD= 0.0025 DELDD= 0.0336 SIGT= 23265.9 SIGX= 11029.1 SIGR= -1336.3 SIGE= 14982.0  
 T=0.1379 X=37.220 XD= 215.48 XD= 641.5 P1= 1164.66 P1D= 0.0 P2= 1164.66 P2D= -21629.3 P3= 1164.66 P3D= 0.0  
 DELA= 0.0077 DELAD=-0.1320 DELD= 0.0022 DELDD= 0.0334 SIGT= 22434.8 SIGX= 10635.1 SIGR= -1293.7 SIGE= 14450.7  
 T=0.1399 X=37.652 XD= 216.73 XD= 607.6 P1= 1121.39 P1D= 0.0 P2= 1121.39 P2D= -21626.6 P3= 1121.39 P3D= 0.0  
 DELA= 0.0074 DELAD=-0.1338 DELD= 0.0021 DELDD= 0.0334 SIGT= 21601.3 SIGX= 10240.0 SIGR= -1121.4 SIGE= 13914.7  
 T=0.1419 X=38.087 XD= 217.91 XD= 573.5 P1= 1074.00 P1D= 0.0 P2= 1078.00 P2D= -2175.5 P3= 1078.00 P3D= 0.0  
 DELA= 0.0071 DELAD=-0.1327 DELD= 0.0020 DELDD= 0.0334 SIGT= 20765.6 SIGX= 9843.8 SIGR= -1078.0 SIGE= 13376.4  
 T=0.1439 X=38.123 XD= 219.02 XD= 539.4 P1= 1034.53 P1D= 0.0 P2= 1034.53 P2D= -21752.6 P3= 1034.53 P3D= 0.0  
 DELA= 0.0068 DELAD=-0.1337 DELD= 0.0019 DELDD= 0.0339 SIGT= 19928.1 SIGX= 9446.8 SIGR= -1034.5 SIGE= 12836.9  
 T=0.1459 X=38.962 XD= 220.06 XD= 505.3 P1= 990.97 P1D= 0.0 P2= 990.97 P2D= -21789.1 P3= 990.97 P3D= 0.0  
 DELA= 0.0065 DELAD=-0.1338 DELD= 0.0018 DELDD= 0.0341 SIGT= 19089.0 SIGX= 9049.0 SIGR= -991.0 SIGE= 12296.4  
 T=0.1479 X=39.403 XD= 221.04 XD= 471.0 P1= 947.35 P1D= 0.0 P2= 947.35 P2D= -21830.9 P3= 947.35 P3D= 0.0  
 DELA= 0.0062 DELAD=-0.1337 DELD= 0.0018 DELDD= 0.0341 SIGT= 18248.7 SIGX= 8650.7 SIGR= -941.3 SIGE= 11755.1  
 T=0.1499 X=39.846 XD= 221.95 XD= 436.4 P1= 903.67 P1D= 0.0 P2= 903.67 P2D= -21827.6 P3= 903.67 P3D= 0.0  
 DELA= 0.0059 DELAD=-0.1345 DELD= 0.0017 DELDD= 0.0341 SIGT= 17407.4 SIGX= 8251.9 SIGR= -903.7 SIGE= 11213.2  
 T=0.1519 X=40.290 XD= 222.77 XD= 374.4 P1= 889.47 P1D= 0.0 P2= 829.28 P2D= -38172.0 P3= 829.28 P3D= 0.0  
 DELA= 0.0055 DELAD=-0.14231 DELD= 0.0015 DELDD= 0.0388 SIGT= 15974.4 SIGX= 7572.6 SIGR= -829.3 SIGE= 10290.1  
 T=0.1539 X=40.736 XD= 223.45 XD= 308.8 P1= 889.47 P1D= 38505.4 P2= 739.23 P2D= -28371.3 P3= 739.23 P3D= 0.0  
 DELA= 0.0049 DELAD=-0.224 DELD= 0.0014 DELDD= 0.0665 SIGT= 14239.7 SIGX= 6750.2 SIGR= -7139.2 SIGE= 9172.7  
 T=0.1559 X=41.193 XD= 224.00 XD= 237.4 P1= 889.47 P1D= 12257.1 P2= 648.01 P2D= -13475.3 P3= 648.01 P3D= 0.0  
 DELA= 0.0043 DELAD=-0.384 DELD= 0.0012 DELDD= 0.0606 SIGT= 12494.2 SIGX= 5922.8 SIGR= -648.6 SIGE= 8048.3  
 T=0.1578 X=41.631 XD= 224.40 XD= 166.9 P1= 889.47 P1D= -8561.8 P2= 558.6 P2D= -17634.1 P3= 558.26 P3D= 0.0  
 DELA= 0.0037 DELAD=-0.1496 DELD= 0.0010 DELDD= 0.0851 SIGT= 10753.7 SIGX= 5097.7 SIGR= -558.3 SIGE= 6927.2  
 T=0.1598 X=42.080 XD= 224.66 XD= 96.2 P1= 889.47 P1D= 25975.6 P2= 468.53 P2D= -3577.8 P3= 468.53 P3D= 0.0  
 DELA= 0.0031 DELD= 0.3844 DELD= 0.0009 DELD= 0.056 SIGT= 9025.3 SIGX= 4278.4 SIGR= -648.5 SIGE= 5813.7  
 T=0.1618 X=42.529 XD= 224.78 XD= 25.7 P1= 889.47 P1D= 41223.7 P2= 378.33 P2D= -7063.2 P3= 378.33 P3D= 0.0  
 DELA= 0.0025 DELAD=-0.2067 DELD= 0.0007 DELDD= 0.0787 SIGT= 7287.8 SIGX= 3454.7 SIGR= -378.3 SIGE= 4694.5

T=0.1638 X=42.974 XD= 215.26 XDD=-12757.3 P1= Aaq,47 PID=-617611.4 P2= 290.43 P2D= 1734.1 P3= 290.43 P3D= 0.0  
 DELA= 0.0019 DELAD=0.0005 DELD=-0.2839 DELD0=-0.0005 DELD0=0.0005 SIGT= 5594.6 SIGAX= 265.01 SIGR= -290.4 SIGE= 3603.8  
 T=0.1658 X=3.357 XD= 153.25 XDD=-57085.0 P1= 889.47 PID=-1214377.0 P2= 215.51 P2D= 18756.5 P3= 215.51 P3D= 0.0  
 DELA= 0.0016 DELAD=0.0004 DELD=0.0004 DELD0=0.0004 SIGT= 4151.4 SIGAX= 1988.0 SIGR= -21.5 SIGE= 216.51 P30= 0.0  
 T=0.1678 X=43.510 XD= 14.75 P0= -9854.5 P1= 889.47 PID=-2006387.0 P2= 188.71 P2D= 59003.5 P3= 188.71 P3D= 0.0  
 DELA= 0.0112 DELAD= 0.0004 DELD=0.0004 DELD0=0.0004 SIGT= 3635.0 SIGAX= 1723.2 SIGR= -188.7 SIGE= 2674.2  
 T=0.1698 X=43.298 XD= 145.17 XD0= -6565.7 P1= 889.47 PID=-88895951.0 P2= 228.59 P2D= 105279.8 P3= 228.59 P3D= 0.0  
 T=0.1714 X=42.818 XD= -284.10 XDD=-35582.8 A P1= Aaq,47 PID=-4171347.0 P2= 208.73 SIGR= -228.6 SIGE= 2836.4  
 DELA= 0.0021 DELAD= 0.4326 DELD=0.0006 DELD0=0.0080 SIGT= 6258.1 SIGAX= 2966.6 SIGR= -324.9 SIGE= 322.88 P3D= 0.0  
 T=0.1738 X=42.191 XD= -337.69 XDD=-19332.7 P1= Aaq,47 PID=-2006387.0 P2= 453.91 P2D= 138239.5 P3= 453.91 P3D= 0.0  
 DELA= 0.0030 DELAD= 0.3523 DELD=0.0008 DELD0=0.01155 SIGT= 4144.9 SIGAX= 453.91 P2D= 135182.1 P3= 453.91 P3D= 0.0  
 T=0.1758 X=41.680 XD= 371.01 XD0= -14370.1 P1= 889.47 PID=-42181813.0 P2= 601.94 P2D= 601.94 P3= 601.94 P3D= 0.0  
 DELA= 0.0039 DELAD= 0.5354 DELD=0.0011 DELD0=0.01144 SIGT= 4403.7 SIGAX= 5496.6 SIGR= -5496.6 SIGE= 601.9 SIGE= 5632.4  
 T=0.1778 X=40.710 XD= -397.93 XD0= -12173.9 P1= Aaq,47 PID=-3742950.0 P2= 765.15 P2D= 133552.3 P3= 765.15 P3D= 0.0  
 DELA= 0.0050 DELAD= 0.4326 DELD=0.0006 DELD0=0.0080 SIGT= 6258.1 SIGAX= 6986.9 SIGR= -765.1 SIGE= 4031.2  
 T=0.1798 X=39.592 XD= -419.90 XD0= -9909.6 P1= 928.60 PID= 928.60 P2D= 40501.3 P3= 453.91 P3D= 0.0  
 DELA= 0.0061 DELAD= 0.2252 DELD=0.0017 DELD0=0.01404 SIGT= 8743.7 SIGAX= 4144.9 SIGR= -453.91 P2D= 135182.1 P3= 453.91 P3D= 0.0  
 T=0.1818 X=39.033 XD= -437.83 XD0= -8105.2 P1= 1012.89 P10= 0.0 P2= 928.60 P2D= 928.60 P3D= 0.0  
 DELA= 0.0067 DELAD= 0.1522 DELD=0.01519 DELD0=0.01144 SIGT= 11595.2 SIGAX= 8479.5 SIGR= -928.6 SIGE= 11522.5  
 T=0.1838 X=38.142 XD= -452.45 XD0= -6581.3 P1= 1101.76 PID= 9249.2 SIGR= -1012.9 SIGE= 7469.2  
 DELA= 0.0072 DELAD= 0.4326 DELD=0.0006 DELD0=0.00144 SIGT= 11595.2 SIGAX= 10060.7 SIGR= -1101.76 SIGE= 12568.5  
 T=0.1858 X=37.224 XD= -464.26 XD0= -5290.7 P1= 1193.92 PID= 1193.92 SIGE= -1101.76 SIGE= 13671.1  
 DELA= 0.0078 DELAD= 0.0649 DELD=0.0022 DELD0=0.0022 DELD0=0.00975 SIGT= 21223.1 SIGAX= 1193.92 P2D= 56017.5 P3= 1193.92 P3D= 0.0  
 T=0.1878 X=36.286 XD= -473.69 XD0= -1287.74 PID= 0.0 P2= 1193.92 P2D= 1193.92 P3D= 0.0  
 DELA= 0.0085 DELAD= 0.4326 DELD=0.00024 DELD0=0.00024 DELD0=0.00571 SIGT= 19551.3 SIGAX= 1281.74 P2D= 4215.2 P3D= 0.0  
 T=0.1898 X=35.331 XD= -481.08 XD0= -3226.1 P1= 1345.42 PID= 1159.0 SIGR= -1287.7 SIGE= 1012.89 P3D= 0.0  
 DELA= 0.0091 DELAD= 0.3076 DELD=0.0026 DELD0=0.0026 DELD0=0.0765 SIGT= 12568.5  
 T=0.1918 X=34.362 XD= -486.51 XD0= -2425.2 P1= 1483.82 PID= 1385.42 P2D= 1101.76 P3D= 0.0  
 DELA= 0.0097 DELAD= 0.1707 DELD=0.0028 DELD0=0.0028 DELD0=0.0945 SIGT= 26687.4 SIGAX= 10060.7 SIGR= -1101.76 P3D= 0.0  
 T=0.1938 X=33.384 XD= -490.82 XD0= -1713.1 P1= 1584.21 PID= 1483.82 P2D= 1101.76 P3D= 0.0  
 DELA= 0.0104 DELAD= 0.4326 DELD=0.00029 DELD0=0.00029 DELD0=0.0557 SIGT= 30516.7 SIGAX= 1354.95 SIGR= -1483.82 P2D= 1101.76 P3D= 0.0  
 T=0.1958 X=32.399 XD= -493.559 XD0= -1090.1 P1= 1686.68 PID= 1159.0 SIGR= -1287.7 SIGE= 15978.9  
 DELA= 0.0111 DELAD= 0.2133 DELD=0.0031 DELD0=0.0031 DELD0=0.0944 SIGT= 26687.4 SIGAX= 1483.82 P2D= 1385.42 P3D= 0.0  
 T=0.1978 X=31.410 XD= -494.95 XD0= -561.2 P1= 1789.91 PID= 1606.68 P2D= 1385.42 P3D= 0.0  
 DELA= 0.0117 DELAD= 0.2412 DELD=0.0033 DELD0=0.0033 DELD0=0.0847 SIGT= 26582.8 SIGAX= 1504.91 P2D= 1606.68 P3D= 0.0  
 T=0.1998 X=30.619 XD= -495.77 XD0= -52.6 P1= 1893.28 PID= 1686.68 SIGR= -1686.7 SIGE= 17191.0  
 DELA= 0.0124 DELAD= 0.4326 DELD=0.00455 DELD0=0.00455 DELD0=0.0557 SIGT= 32490.4 SIGAX= 1789.11 P2D= 1686.68 SIGR= -1686.7 SIGE= 20929.1  
 T=0.2018 X=29.427 XD= -495.42 XD0= -383.7 P1= 1998.27 PID= 1686.68 P2D= 1686.68 P3D= 0.0  
 DELA= 0.0131 DELAD= 0.2068 DELD=0.0037 DELD0=0.0037 DELD0=0.0983 SIGT= 34922.7 SIGAX= 18247.2 SIGR= -1998.3 SIGE= 24795.5  
 T=0.2038 X=28.437 XD= -494.926 XD0= -776.4 P1= 2102.97 PID= 0.0 P2= 2102.97 P2D= 52816.6 P3= 2102.97 P3D= 0.0  
 DELA= 0.0138 DELAD= 0.2312 DELD=0.0043 DELD0=0.0043 DELD0=0.0963 SIGT= 44590.6 SIGAX= 21137.9 SIGR= -2314.63 P3D= 0.0  
 T=0.2058 X=27.450 XD= -494.923.32 XD0= -1133.0 P1= 2420.26 PID= 0.0 P2= 2420.26 P2D= 53741.8 P3= 28723.6 P3D= 0.0  
 DELA= 0.0145 DELAD= 0.4305 DELD=0.0035 DELD0=0.0041 DELD0=0.0680 SIGT= 36470.2 SIGAX= 19203.3 SIGR= -2103.0 SIGE= 22200.1  
 T=0.2078 X=26.68 XD= -489.71 XD0= 1458.6 P1= 2314.83 PID= 17288.5 SIGR= -1893.3 SIGE= 1893.3 SIGE= 1893.3 SIGE= 228 P3D= 0.0  
 DELA= 0.0152 DELAD= 0.2312 DELD=0.0043 DELD0=0.0043 DELD0=0.0983 SIGT= 44590.6 SIGAX= 1982.66 P2D= 57656.3 P3= 1998.3 SIGE= 23192.7  
 T=0.2098 X=25.491 XD= -494.926 XD0= -776.4 P1= 2102.97 PID= 0.0 P2= 2102.97 P2D= 52816.6 P3= 2102.97 P3D= 0.0  
 DELA= 0.0159 DELAD= 0.3078 DELD=0.0045 DELD0=0.0045 DELD0=0.0944 SIGT= 466621.0 SIGAX= 19203.3 SIGR= -2103.0 SIGE= 22200.1  
 T=0.2118 X=24.521 XD= -492.32 XD0= -1133.0 P1= 2208.80 PID= 0.0 P2= 2208.80 P2D= 48034.4 P3= 2208.80 P3D= 0.0  
 DELA= 0.0166 DELAD= 0.4315 DELD=0.0047 DELD0=0.0047 DELD0=0.0690 SIGT= 42568.0 SIGAX= 17288.5 SIGR= -1893.3 SIGE= 23192.7  
 T=0.2138 X=23.560 XD= -477.34 XD0= 2291.0 P1= 2631.99 PID= 1998.27 P2D= 1998.27 P2D= 57656.3 P3= 1998.27 P3D= 0.0  
 DELA= 0.0173 DELAD= 0.2312 DELD=0.0049 DELD0=0.0049 DELD0=0.0906 SIGT= 50700.0 SIGAX= 2034.0 SIGR= -2632.0 SIGE= 32659.0  
 T=0.2258 X=22.608 XD= -477.35 XD0= 2531.1 P1= 2736.91 PID= 0.0 P2= 2736.91 P2D= 55111.2 P3= 32659.0  
 DELA= 0.0180 DELAD= 0.2678 DELD=0.0051 DELD0=0.0051 DELD0=0.0894 SIGT= 52720.9 SIGAX= 22010.6 SIGR= -2420.26 P3D= 0.0  
 T=0.2178 X=21.656 XD= -468.21 XD0= -277.2 P1= 2841.55 PID= 0.0 P2= 2526.16 P2D= 48318.9 P3= 2526.16 P3D= 0.0  
 DELA= 0.0167 DELAD= 0.3954 DELD=0.0057 DELD0=0.0053 DELD0=0.0705 SIGT= 54736.8 SIGAX= 23067.6 SIGR= -2526.2 SIGE= 3134.63 P3D= 0.0  
 T=0.2198 X=20.735 XD= -467.47 XD0= 2910.5 P1= 2945.99 PID= 2594.76 SIGR= -2641.6 SIGE= 35259.4  
 DELA= 0.0193 DELAD= 0.3322 DELD=0.0055 DELD0=0.0055 DELD0=0.0973 SIGT= 56748.5 SIGAX= 26901.3 SIGR= -2945.99 SIGR= 32659.0  
 T=0.2218 X=19.816 XD= -465.54 XD0= 2537.3 P1= 3014.76 PID= 0.0 P2= 3014.76 P2D= 55111.2 P3= 36555.2  
 DELA= 0.0194 DELAD= 0.5388 DELD=0.0056 DELD0=0.0056 DELD0=0.0894 SIGT= 58073.3 SIGAX= 24992.0 SIGR= -236.9 SIGE= 33900.8  
 T=0.2238 X=18.919 XD= -439.12 XD0= 10016.1 P1= 3066.78 PID= 0.0 P2= 3066.78 P2D= 48658.1 P3= 2841.55 P3D= 0.0  
 DELA= 0.0202 DELAD= 0.6912 DELD=0.0057 DELD0=0.0657 SIGT= 59075.3 SIGAX= 26308.7 P3= 5619.99 P3D= 359613.4  
 T=0.2258 X=18.662 XD= -417.79 XD0= 11052.6 P1= 3157.68 PID= 28046.3 SIGR= -3066.6 SIGE= 36054.1  
 DELA= 0.0207 DELAD= 0.7949 DELD=0.0059 DELD0=0.0066 SIGT= 60826.2 SIGAX= 2157.68 P2D= 19764.0 P3= 6023.12 P3D= 103130.0  
 T=0.2278 X=17.248 XD= -395.35 XD0= 11315.6 P1= 3251.29 PID= 0.0 P2= 3251.29 P2D= 61214.1 P3= 6148.50 P3D= 3768.1

DELA= 0.0213 DELAD= 0.0262 DELO= 0.0060 DELDD= 0.1213 SIGT= 62629.6 SIGAX= -29689.2 SIGR= -3251.3 SIGE= 40343.6  
 T=0.2298 X=16.480 XD= -312.65 XDO= 11354.3 P1= 3339.42 P10= 0.0 P2= 3339.42 P2D= 69233.9 P3= 6191.90 P3D= 11659.1  
 DELA= 0.0219 DELAD= 0.0228 DELO= 0.0062 DELDD= -0.1062 SIGT= 64327.1 SIGAX= 30493.9 SIGR= -3339.4 SIGE= 41437.1  
 T=0.2318 X=15.757 XD= -350.00 XDO= 11286.5 P1= 3423.39 P10= 0.0 P2= 3423.39 P2D= 3465.0 P3= 6196.66 P3D= -4990.6  
 DELA= 0.0225 DELAD= 0.0238 DELO= 0.0064 DELDD= -0.0592 SIGT= 65944.6 SIGAX= 3120.6 SIGR= -3423.4 SIGE= 42479.0  
 T=0.2338 X=15.079 XD= -327.56 XDO= 11142.4 P1= 3505.05 P10= 0.0 P2= 3505.05 P2D= 15280.7 P3= 6173.00 P3D= -17383.1  
 DELA= 0.0230 DELAD= 0.0240 DELD= 0.0065 DELDD= 0.0177 SIGT= 67517.7 SIGAX= 3336.4 SIGR= -3505.1 SIGE= 43492.3  
 T=0.2358 X=14.446 XD= -305.47 XDO= 10935.7 P1= 3582.83 P10= 0.0 P2= 3582.83 P2D= 3400.3 P3= 6124.32 P3D= -27847.4  
 DELA= 0.0235 DELAD= 0.0261 DELD= 0.0067 DELDD= -0.0552 SIGT= 69016.0 SIGAX= 32716.6 SIGR= -3582.8 SIGE= 44447.5  
 T=0.2378 X=13.857 XD= -282.85 XDO= 10674.3 P1= 3654.51 P10= 0.0 P2= 3654.51 P2D= 54579.9 P3= 6053.14 P3D= -38274.4  
 DELA= 0.0240 DELAD= 0.0263 DELD= 0.0068 DELDD= -0.1182 SIGT= 70396.7 SIGAX= 33371.2 SIGR= -3654.5 P3= 4536.9  
 T=0.2394 X=13.311 XD= -262.81 XDO= 10363.1 P1= 3720.54 P10= 0.0 P2= 3720.54 P2D= 42336.1 P3= 5962.33 P3D= -48229.3  
 DELA= 0.0244 DELAD= 0.0269 DELD= 0.0069 DELDD= -0.0795 SIGT= 71668.6 SIGAX= 33974.1 SIGR= -3720.5 SIGE= 46166.2  
 T=0.2418 X=12.805 XD= -224.24 XDO= 10000.6 P1= 3742.94 P10= 0.0 P2= 3742.84 P2D= 17612.2 P3= 5855.07 P3D= -55875.8  
 DELA= 0.0249 DELAD= 0.0280 DELD= 0.0070 DELDD= -0.0049 SIGT= 72868.7 SIGAX= 34543.0 SIGR= -3782.8 SIGE= 4639.2  
 DELA= 0.0252 DELAD= 0.0281 XDO= 9624.7 P1= 3841.88 P10= 0.0 P2= 3841.88 P2D= 13542.3 P3= 5734.77 P3D= -60593.8  
 T=0.2458 X=11.914 XD= -203.97 XDO= 9220.3 P1= 3890.61 SIGT= 74006.1 SIGAX= 35082.1 SIGR= -3841.9 SIGE= 47671.9  
 DELA= 0.0256 DELAD= 0.0291 DELD= 0.0073 DELDD= -0.0543 SIGT= 75056.6 SIGAX= 35580.1 SIGR= -3896.4 SIGE= 48148.6  
 DELA= 0.0259 DELAD= 0.0292 DELD= 0.0073 DELDD= -0.0823 SIGT= 76003.6 SIGAX= 36029.1 SIGR= -3945.6 SIGE= 48958.7  
 T=0.2498 X=11.169 XD= -164.77 XDO= 8377.9 P1= 3990.09 P10= 0.0 P2= 3990.09 P2D= 26368.0 P3= 5331.70 P3D= -66807.1  
 DELA= 0.0262 DELAD= 0.0349 DELD= 0.0074 DELDD= -0.0474 SIGT= 76861.0 SIGAX= 36435.5 SIGR= -3999.1 SIGE= 49510.9  
 T=0.2518 X=10.848 XD= -152.42 XDO= 8151.42 P1= 4031.17 P10= 0.0 P2= 4031.17 P2D= 10201.4 P3= 5605.04 P3D= -63510.8  
 DELA= 0.0265 DELAD= 0.0292 DELD= 0.0075 DELDD= -0.0004 SIGT= 77652.4 SIGAX= 36810.6 SIGR= -4019.2 SIGE= 50020.7  
 T=0.2538 X=10.559 XD= -185.76 XDO= 8805.4 P1= 3945.58 P10= 0.0 P2= 3945.58 P2D= 37889.5 P3= 5469.52 P3D= -65706.3  
 DELA= 0.0267 DELAD= 0.0294 DELD= 0.0076 DELDD= -0.0073 SIGT= 78003.6 SIGAX= 36029.1 SIGR= -3945.6 SIGE= 48958.7  
 T=0.2557 X=10.300 XD= -122.12 XDO= 7919.7 P1= 4103.09 P10= 0.0 P2= 4103.09 P2D= 1793.09 P3= 5331.70 P3D= -66807.1  
 DELA= 0.0269 DELAD= 0.0562 DELD= 0.0076 DELDD= -0.0336 SIGT= 79037.6 SIGAX= 37467.3 SIGR= -4103.1 SIGE= 50913.0  
 T=0.2577 X=10.070 XD= -114.08 XDO= 6848.4 P1= 4122.76 P10= 0.0 P2= 4132.76 P2D= 23639.1 P3= 4815.22 P3D= -54161.2  
 DELA= 0.0271 DELAD= 0.0292 DELD= 0.0077 DELDD= -0.0530 SIGT= 79669.2 SIGAX= 37738.3 SIGR= -4132.8 SIGE= 51281.3  
 T=0.2597 X= 9.867 XD= -94.72 XDO= 6526.2 P1= 4158.62 P10= 0.0 P2= 4158.62 P2D= 16651.2 P3= 4705.62 P3D= -626681.8  
 DELA= 0.0273 DELAD= 0.0338 DELD= -0.0077 DELDD= -0.0025 SIGT= 76382.3 SIGAX= 37156.7 SIGR= -4060.1 SIGE= 50490.9  
 T=0.2617 X= 9.691 XD= -81.97 XDO= 6235.7 P1= 4191.47 P10= 0.0 P2= 4181.47 P2D= 5411.2 P3= 4606.78 P3D= -58426.8  
 DELA= 0.0275 DELAD= 0.0275 XDO= 6078.0 P1= 4007.91 SIGT= 80547.5 SIGAX= 38183.1 SIGR= -4181.5 SIGE= 51885.7  
 T=0.2637 X= 9.539 XD= -69.77 XDO= 5979.5 P1= 4201.60 P10= 0.0 P2= 4201.60 P2D= 1915.8 P3= 4510.50 P3D= -54161.2  
 DELA= 0.0276 DELAD= 0.0295 DELD= 0.0077 DELDD= -0.0530 SIGT= 79669.2 SIGAX= 38366.9 SIGR= -4201.6 SIGE= 52135.5  
 T=0.2657 X= 9.411 XD= -58.04 XDO= 5759.9 P1= 4248.65 P10= 0.0 P2= 4248.65 P2D= 7372.9 P3= 4444.32 P3D= -49872.9  
 DELA= 0.0277 DELAD= 0.0501 DELD= -0.0051 SIGT= 80107.4 SIGAX= 37974.4 SIGR= -4158.6 SIGE= 51602.2  
 T=0.2677 X= 9.307 XD= -44.65.72 XDO= 5623.7 P1= 4191.47 P10= 0.0 P2= 4181.47 P2D= 5411.2 P3= 4606.78 P3D= -44730.1  
 DELA= 0.0278 DELAD= 0.0191 DELD= 0.0078 DELDD= -0.0078 SIGT= 80547.5 SIGAX= 38183.1 SIGR= -4181.5 SIGE= 50913.0  
 T=0.2697 X= 9.224 XD= -35.71 XDO= 5433.1 P1= 4242.75 P10= 0.0 P2= 4242.75 P2D= 1915.8 P3= 4510.50 P3D= -38248.8  
 DELA= 0.0279 DELAD= 0.0486 DELD= 0.0079 DELDD= -0.0079 SIGT= 80935.3 SIGAX= 38366.9 SIGR= -4201.6 SIGE= 52135.5  
 T=0.2717 X= 9.163 XD= -24.96 XDO= 5325.7 P1= 4250.46 P10= 0.0 P2= 4250.46 P2D= 1194.6 P3= 4294.59 P3D= -31179.8  
 DELA= 0.0279 DELAD= 0.0594 DELD= -0.0079 DELDD= -0.0119 SIGT= 81263.1 SIGAX= 38522.6 SIGR= -4218.7 SIGE= 52347.0  
 T=0.2737 X= 9.124 XD= -14.39 XDO= 5577.8 P1= 4232.31 P10= 0.0 P2= 4232.31 P2D= 1191.8 P3= 4308.62 P3D= -24717.6  
 DELA= 0.0279 DELAD= 0.1005 DELD= 0.0079 DELDD= -0.0289 SIGT= 81526.8 SIGAX= 38647.3 SIGR= -4232.3 SIGE= 52516.5  
 T=0.2757 X= 9.106 XD= -3.92 XDO= 5224.8 P1= 4258.27 P10= 0.0 P2= 4258.27 P2D= 8801.1 P3= 4331.65 P3D= -19273.0  
 DELA= 0.0280 DELAD= 0.0270 XDO= 5026.4 P1= 4253.16 P10= 0.0 P2= 4253.16 P2D= 794.1 P3= 4259.36 P3D= -1981.9  
 T=0.2777 X= 9.107 XD= -3.80 XDO= 3039.7 P1= 4258.16 P10= 0.0 P2= 4258.16 P2D= 1194.6 P3= 4294.59 P3D= -14137.5  
 DELA= 0.0280 DELAD= 0.0651 DELD= -0.0051 SIGT= 81876.5 SIGAX= 38813.1 SIGR= -4250.5 SIGE= 52741.7  
 T=0.2797 X= 9.120 XD= -27.74 XDO= 3027.3 P1= 4256.32 P10= 0.0 P2= 4256.32 P2D= 2906.9 P3= 4270.48 P3D= -8045.8  
 DELA= 0.0280 DELAD= 0.0059 DELD= -0.0059 SIGT= 81977.4 SIGAX= 38809.9 SIGR= -4255.7 SIGE= 52886.8  
 T=0.2817 X= 9.146 XD= -15.80 XDO= 3001.3 P1= 4253.16 P10= 0.0 P2= 4253.16 P2D= 794.1 P3= 4259.36 P3D= -7635.3  
 DELA= 0.0279 DELAD= 0.0060 DELD= -0.0270 XDO= 2962.3 P1= 4249.82 P10= 0.0 P2= 4249.82 P2D= 4253.2 SIGE= 52775.2  
 T=0.2837 X= 9.184 XD= -3.86 XDO= 2787.1 P1= 4222.55 P10= 0.0 P2= 4227.55 P2D= 2859.3 P3= 4257.13 P3D= 1835.3  
 DELA= 0.0279 DELAD= 0.0355 DELD= -0.0079 DELDD= -0.0125 SIGT= 81844.8 SIGAX= 38833.4 SIGR= -4258.2 SIGE= 52721.3  
 T=0.2857 X= 9.233 XD= -9.87 XDO= 2912.6 P1= 4243.18 P10= 0.0 P2= 4243.18 P2D= 1685.5 P3= 4249.93 P3D= -2755.8  
 DELA= 0.0279 DELAD= 0.0057 DELD= -0.0079 DELDD= -0.0062 SIGT= 81989.4 SIGAX= 38866.6 SIGR= -4255.7 SIGE= 52884.5  
 T=0.2877 X= 9.265 XD= -33.51 XDO= 2851.8 P1= 4236.10 P10= 0.0 P2= 4236.10 P2D= 2356.5 P3= 4236.94 P3D= -7635.3  
 DELA= 0.0278 DELAD= 0.0523 DELD= -0.0079 DELDD= -0.0015 SIGT= 81928.4 SIGAX= 38848.3 SIGR= -4253.2 SIGE= 52886.6  
 T=0.2897 X= 9.367 XD= -39.15 XDO= 2787.1 P1= 4222.55 P10= 0.0 P2= 4227.55 P2D= 2603.9 P3= 4133.50 P3D= -15803.2  
 DELA= 0.0278 DELAD= 0.0642 DELD= -0.0079 DELDD= -0.0006 SIGT= 81435.2 SIGAX= 38863.9 SIGR= -4227.6 SIGE= 52457.5  
 T=0.2917 X= 9.451 XD= -44.65 XDO= 2713.9 P1= 4217.71 P10= 0.0 P2= 4217.71 P2D= 5189.7 P3= 4097.54 P3D= -17814.0  
 DELA= 0.0277 DELAD= 0.0310 DELD= -0.0078 DELDD= -0.0019 SIGT= 81245.6 SIGAX= 38514.0 SIGR= -4217.7 SIGE= 52335.4

T=0.2937 X= 9.546 XD= 49.99 XDD= 2635.8 P1= 4206.71 P1D= 0.0 P2= 4206.71 P2D= 7459.6 P3= 4058.81 P3D= -19100.5  
 DELA= 0.0276 DELAD=-0.0064 DELD= 0.0146 SIGT= 81033.7 SIGX= 4206.7 SIGR= -4206.7 SIGE= 52198.8  
 DELA= 9.651 XD= 55.18 XDD= 2554.3 P1= 4194.54 P1D= 0.0 P2= 4194.3 SIGR= -4194.3 SIGE= 52198.8  
 DELA= 0.0275 DELAD=-0.0196 DELD= -0.0178 DELD0= 0.0135 SIGT= 80799.2 SIGAX= 4194.5 P2D= -7466.6 P3= 4017.9 P3D= -19824.0  
 T=0.2977 X= 9.766 XD= 60.21 XDD= 2471.3 P1= 4181.12 P1D= 0.0 P2= 4181.2 P2D= 4194.5 SIGR= -4194.5 SIGE= 52047.8  
 DELA= 0.0275 DELAD=-0.0549 DELD= 0.0078 DELD0= 0.0091 SIGT= 80540.7 SIGAX= 4181.7 P2D= -6372.6 P3= 3975.48 P3D= -20402.3  
 DELA= 0.0274 DELAD=-0.0715 DELD= -0.0078 DELD0= 0.0082 SIGT= 80258.1 SIGAX= 4181.9 SIGR= -4181.9 SIGE= 51881.3  
 T=0.3017 X=10.026 XD= 69.75 XDD= 2304.7 P1= 4150.60 P1D= 0.0 P2= 4166.45 P2D= -6232.2 P3= 3932.04 P3D= -20947.9  
 DELA= 0.0273 DELAD=-0.0559 DELD= 0.0077 DELD0= 0.0119 SIGT= 79952.9 SIGAX= 38045.9 SIGR= -4166.4 SIGE= 51699.2  
 T=0.3037 X=10.171 XD= 74.28 XDD= 2223.1 P1= 4133.73 P1D= 0.0 P2= 4133.73 P2D= -7818.2 P3= 3888.06 P3D= -21342.3  
 DELA= 0.0272 DELAD=-0.0350 DELD= 0.0077 DELD0= 0.0166 SIGT= 79627.9 SIGAX= 37901.2 SIGR= -4150.6 SIGE= 51522.6  
 T=0.1957 X=10.321 XD= 76.48 XD0= 2367.7 P1= 4166.45 P1D= 0.0 P2= 4166.45 P2D= -6313.7 SIGE= 51881.3  
 DELA= 0.0270 DELAD=-0.0416 DELD= 0.0071 DELD0= 0.0169 SIGT= 79263.6 SIGAX= 38045.9 SIGR= -4166.4 SIGE= 51699.2  
 DELA= 0.0269 DELAD=-0.0646 DELD= 0.0076 DELD0= 0.0149 SIGT= 78919.3 SIGAX= 37583.9 SIGR= -4115.9 SIGE= 51071.5  
 T=0.3097 X=10.655 XD= 86.91 XD0= 1995.0 P1= 4011.71 P1D= 0.0 P2= 4096.95 P2D= -9466.7 P3= 3756.72 P3D= -20878.6  
 DELA= 0.0268 DELAD=-0.0771 DELD= 0.0076 DELD0= 0.0136 SIGT= 78535.6 SIGAX= 3741.2 SIGR= -4150.6 SIGE= 50836.8  
 T=0.3117 X=10.832 XD= 90.43 XD0= 1925.9 P1= 4056.13 P1D= 0.0 P2= 4077.02 P2D= -9413.9 P3= 3843.99 P3D= -21412.4  
 DELA= 0.0266 DELAD=-0.0707 DELD= 0.0071 DELD0= 0.0161 SIGT= 78130.0 SIGAX= 4115.86 P2D= -9433.7 SIGE= 51293.2  
 T=0.3137 X=11.018 XD= 94.61 XD0= 2067.7 P1= 4096.95 P1D= 0.0 P2= 4056.13 P2D= -9995.9 SIGE= 53800.12 P3D= -21194.7  
 DELA= 0.0265 DELAD=-0.0587 DELD= 0.0075 DELD0= 0.0148 SIGT= 77713.4 SIGAX= 37583.9 SIGR= -4115.9 SIGE= 51071.5  
 T=0.3157 X=11.221 XD= 105.24 XD0= 1799.2 P1= 4011.71 P1D= 0.0 P2= 4034.35 P2D= -1623.4 P3= 3631.21 P3D= -19898.8  
 DELA= 0.0263 DELAD=-0.0598 DELD= 0.0075 DELD0= 0.0199 SIGT= 77277.5 SIGAX= 36830.6 SIGR= -4096.9 SIGE= 50060.1  
 T=0.3177 X=11.411 XD= 101.81 XD0= 1741.8 P1= 3948.23 P1D= 0.0 P2= 4011.71 P2D= -12132.4 P3= 3591.20 P3D= -19417.3  
 DELA= 0.0262 DELAD=-0.0744 DELD= 0.0074 DELD0= 0.0183 SIGT= 76825.1 SIGAX= 3698.23 P2D= -1073.0 SIGE= 49779.2  
 T=0.3197 X=11.618 XD= 105.24 XD0= 1648.2 P1= 3963.88 P1D= 0.0 P2= 3988.23 P2D= -1191.9 P3= 3352.16 P3D= -18925.2  
 DELA= 0.0260 DELAD=-0.0861 DELD= 0.0075 DELD0= 0.0174 SIGT= 76356.0 SIGAX= 3619.6 SIGR= -4183.8 P3= 3398.6 P3D= -18514.5  
 T=0.3217 X=11.832 XD= 104.56 XD0= 117.98 P1= 3938.71 P1D= 0.0 P2= 3988.71 P2D= -1251.1 P3= 3476.88 P3D= -18514.5  
 DELA= 0.0259 DELAD=-0.0032 DELD= 0.0073 DELD0= 0.0188 SIGT= 75871.2 SIGAX= 3596.3 SIGR= -4011.7 SIGE= 49779.2  
 T=0.3237 X=12.052 XD= 111.79 XD0= 1591.7 P1= 3912.79 P1D= 0.0 P2= 3912.79 P2D= -13468.2 P3= 3440.57 P3D= -17798.9  
 DELA= 0.0252 DELAD=-0.0742 DELD= 0.0073 DELD0= 0.0211 SIGT= 75372.0 SIGAX= 35729.7 SIGR= -35729.7 SIGE= 49487.8  
 T=0.3257 X=12.279 XD= 114.93 XD0= 1548.5 P1= 3886.17 P1D= 0.0 P2= 3886.17 P2D= -13859.3 P3= 3405.05 P3D= -17413.1  
 DELA= 0.0255 DELAD=-0.0762 DELD= 0.0072 DELD0= 0.0221 SIGT= 74859.1 SIGAX= 3619.6 SIGR= -3933.9 SIGE= 49185.7  
 T=0.3277 X=12.512 XD= 117.98 XD0= 1508.3 P1= 3858.62 P1D= 0.0 P2= 3858.62 P2D= -1374.8 P3= 3476.88 P3D= -16159.4  
 DELA= 0.0253 DELAD=-0.0747 DELD= 0.0072 DELD0= 0.0213 SIGT= 74332.2 SIGAX= 35236.8 SIGR= -3586.8 SIGE= 498873.4  
 T=0.3297 X=12.751 XD= 120.96 XD0= 1741.7 P1= 3912.79 P1D= 0.0 P2= 3912.79 P2D= -13468.2 P3= 3440.57 P3D= -17798.9  
 DELA= 0.0252 DELAD=-0.0951 DELD= 0.0073 DELD0= 0.0214 SIGT= 73791.5 SIGAX= 3498.4 SIGR= -35729.7 SIGE= 48551.8  
 T=0.3317 X=12.95 XD= 132.87 XD0= 1436.4 P1= 3801.98 P1D= 0.0 P2= 3801.98 P2D= -14362.2 P3= 3302.44 P3D= -17413.1  
 DELA= 0.0250 DELAD=-0.0762 DELD= 0.0072 DELD0= 0.0221 SIGT= 73237.4 SIGAX= 3486.5 SIGR= -3933.9 SIGE= 48221.4  
 T=0.3337 X=13.246 XD= 126.70 XD0= 1404.0 P1= 3772.58 P1D= 0.0 P2= 3772.58 P2D= -1374.8 P3= 3476.88 P3D= -16159.4  
 DELA= 0.0253 DELAD=-0.0660 DELD= 0.0070 DELD0= 0.0240 SIGT= 72671.1 SIGAX= 3449.3 SIGR= -3586.8 SIGE= 498873.4  
 T=0.3357 X=13.502 XD= 129.49 XD0= 1374.7 P1= 3838.75 P1D= 0.0 P2= 3830.75 P2D= -13798.3 P3= 3336.11 P3D= -16790.9  
 DELA= 0.0246 DELAD=-0.0888 DELD= 0.0070 DELD0= 0.0244 SIGT= 72093.0 SIGAX= 3498.4 SIGR= -3830.7 SIGE= 48551.8  
 T=0.3377 X=13.764 XD= 132.20 XD0= 1355.7 P1= 3711.94 P1D= 0.0 P2= 3801.98 P2D= -14362.2 P3= 3302.44 P3D= -16182.0  
 DELA= 0.0244 DELAD=-0.0923 DELD= 0.0071 DELD0= 0.0222 SIGT= 71502.9 SIGAX= 3471.8 SIGR= -3802.0 SIGE= 47176.7  
 T=0.3397 X=14.031 XD= 134.86 XD0= 1319.3 P1= 3660.69 P1D= 0.0 P2= 3711.94 SIGR= -3802.0 SIGE= 47176.7  
 DELA= 0.0242 DELAD=-0.1019 DELD= 0.0069 DELD0= 0.0236 SIGT= 70901.0 SIGAX= 3319.5 SIGR= -3711.9 SIGE= 4605.5  
 T=0.3417 X=14.303 XD= 1374.7 P1= 3648.85 P1D= 0.0 P2= 3680.69 P2D= -15500.0 P3= 3269.48 P3D= -17022.6  
 DELA= 0.0240 DELAD=-0.0992 DELD= -0.0064 DELD0= 0.0221 SIGT= 70287.6 SIGAX= 3449.3 SIGR= -3602.7 SIGE= 47882.0  
 T=0.3437 X=14.581 XD= 140.04 XD0= 1270.9 P1= 3616.46 P1D= 0.0 P2= 3742.57 P2D= -15370.7 P3= 3326.84 P3D= -16790.9  
 DELA= 0.0238 DELAD=-0.0960 DELD= 0.0069 DELD0= 0.0232 SIGT= 69663.7 SIGAX= 3475.2 SIGR= -3747.6 SIGE= 47533.7  
 T=0.3457 X=14.863 XD= 142.56 XD0= 1246.7 P1= 3543.53 P1D= 0.0 P2= 3711.94 SIGR= -3802.0 SIGE= 47176.7  
 DELA= 0.0235 DELAD=-0.0994 DELD= 0.0069 DELD0= 0.0261 SIGT= 69029.4 SIGAX= 3319.5 SIGR= -3711.9 SIGE= 4605.5  
 T=0.3477 X=15.151 XD= 145.03 XD0= 1227.6 P1= 3550.06 P1D= 0.0 P2= 3648.85 P2D= -16000.9 P3= 3172.56 P3D= -15912.8  
 DELA= 0.0233 DELAD=-0.1060 DELD= 0.0064 DELD0= 0.0221 SIGT= 68384.7 SIGAX= 3417.3 SIGR= -16726.3 P3= 3140.80 P3D= -15857.0  
 T=0.3497 X=15.443 XD= 147.47 XD0= 1207.4 P1= 3516.05 P1D= 0.0 P2= 3516.05 P2D= -16542.6 SIGE= 45276.6  
 DELA= 0.0231 DELAD=-0.1089 DELD= 0.0065 DELD0= 0.0230 SIGT= 667729.6 SIGAX= 33023.7 SIGR= -1616.5 SIGE= 44050.8  
 T=0.3516 X=15.741 XD= 149.86 XD0= 1148.1 P1= 3491.54 P1D= 0.0 P2= 3583.53 P2D= -1671.5 P3= 3014.89 P3D= -15800.9  
 DELA= 0.0229 DELAD=-0.1068 DELD= 0.0065 DELD0= 0.0261 SIGT= 67064.7 SIGAX= 3411.54 P2D= -17362.0 P3= 2983.47 P3D= -15748.7  
 T=0.3536 X=16.043 XD= 152.22 XD0= 1160.4 P1= 3446.53 P1D= 0.0 P2= 3446.53 P2D= -17426.3 P3= 304.30 P3D= -15729.1  
 DELA= 0.0226 DELAD=-0.1059 XD= 154.54 XD0= 1066.0 P1= 3446.53 P1D= 0.0 P2= 3446.53 P2D= -17426.3 P3= 2951.99 P3D= -15840.6  
 T=0.3556 X=16.349 XD= 151.3 P1= 3411.3 P1D= 0.0 P2= 3411.05 P2D= -17466.2 SIGE= 42766.2  
 DELA= 0.0224 DELAD=-0.1073 DELD= 0.0063 DELD0= 0.0285 SIGT= 65707.0 SIGAX= 3414.0 P2D= -17965.1 P3= 2920.41 P3D= -15809.2  
 T=0.3576 X=16.660 XD= 156.82 XD0= 1133.8 P1= 3335.10 P1D= 0.0 P2= 3351.50 P2D= -17997.6 P3= 2888.59 P3D= -15959.7

```

DELA= 0.0222 DELAD=-0.1130 DELD=-0.0063 DELDD= 0.0276 SIGT= 65014.4 SIGAX=
T=0.3596 X=16.976 XD= 159.07 XDD= 1116.6 P1= 333A.67 P1D= 0.0 P2=
DELA= 0.0219 DELAD=-0.1144 DELD= 0.0062 DELDD= 0.0285 SIGT= 64312.7 SIGAX=
T=0.3616 X=17.296 XD= 161.29 XDD= 1099.8 P1= 3301.90 P1D= 0.0 P2=
DELA= 0.0217 DELAD=-0.1128 DELD= 0.0061 DELDD= 0.0292 SIGT= 63602.4 SIGAX=
T=0.3636 X=17.620 XD= 163.47 XDD= 1033.3 P1= 3264.50 P1D= 0.0 P2=
DELA= 0.0214 DELAD=-0.1130 DELD= 0.0061 DELDD= 0.0293 SIGT= 62883.9 SIGAX=
T=0.3656 X=17.949 XD= 165.62 XDD= 1061.0 P1= 3226.78 P1D= 0.0 P2=
DELA= 0.0212 DELAD=-0.1174 DELD= 0.0060 DELDD= 0.0293 SIGT= 62157.3 SIGAX=
T=0.3676 X=18.282 XD= 167.74 XDD= 1050.8 P1= 31AA.63 P1D= 0.0 P2=
DELA= 0.0209 DELAD=-0.1205 DELD= 0.0059 DELDD= 0.0294 SIGT= 61422.5 SIGAX=
T=0.3696 X=18.619 XD= 169.82 XDD= 1034.8 P1= 3150.07 P1D= 0.0 P2=
DELA= 0.0207 DELAD=-0.1196 DELD= 0.0059 DELDD= 0.0302 SIGT= 60679.8 SIGAX=
T=0.3714 X=18.961 XD= 171.87 XDD= 1018.8 P1= 3111.13 P1D= 0.0 P2=
DELA= 0.0204 DELAD=-0.1176 DELD= 0.0058 DELDD= 0.0309 SIGT= 59929.7 SIGAX=
T=0.3736 X=19.306 XD= 173.90 XDD= 1002.8 P1= 3071.82 P1D= 0.0 P2=
DELA= 0.0202 DELAD=-0.1212 DELD= 0.0057 DELDD= 0.0305 SIGT= 59172.4 SIGAX=
T=0.3756 X=19.656 XD= 175.88 XDD= 986.8 P1= 3032.13 P1D= 0.0 P2=
DELA= 0.0199 DELAD=-0.1240 DELD= 0.0056 DELDD= 0.0310 SIGT= 5A407.7 SIGAX=
EF= -348.67212
EC= -78452.3750

```

## DISTRIBUTION LIST

Commander  
US Army Armament Research and  
Development Command  
ATTN: DRDAR-TSS (5)  
DRDAR-LCW (25)  
Dover, NJ 07801

Defense Documentation Center (12)  
Cameron Station  
Alexandria, VA 22314

US Army TRADOC Systems  
Analysis Activity  
ATTN: ATAA-SL (Tech Lib)  
White Sands Missile Range, NM 88002

Commander  
US Army Armament Materiel and  
Readiness Command  
ATTN: DRSAR-LEP-L  
Rock Island, IL 61299

US Army Materiel Systems  
Analysis Activity  
ATTN: DRXSY-MP  
Aberdeen Proving Ground, MD 21005

Weapon System Concept Team/CSL  
ATTN: DRDAR-ACW  
Aberdeen Proving Ground, MD 21010

Technical Library  
ATTN: DRDAR-CLJ-L  
Aberdeen Proving Ground, MD 21005

Technical Library  
ATTN: DRDAR-TSB-S  
Aberdeen Proving Ground, MD 21010

Technical Library  
ATTN: DRDAR-LCB-TL  
Benet Weapons Laboratory  
Watervliet, NY 12189