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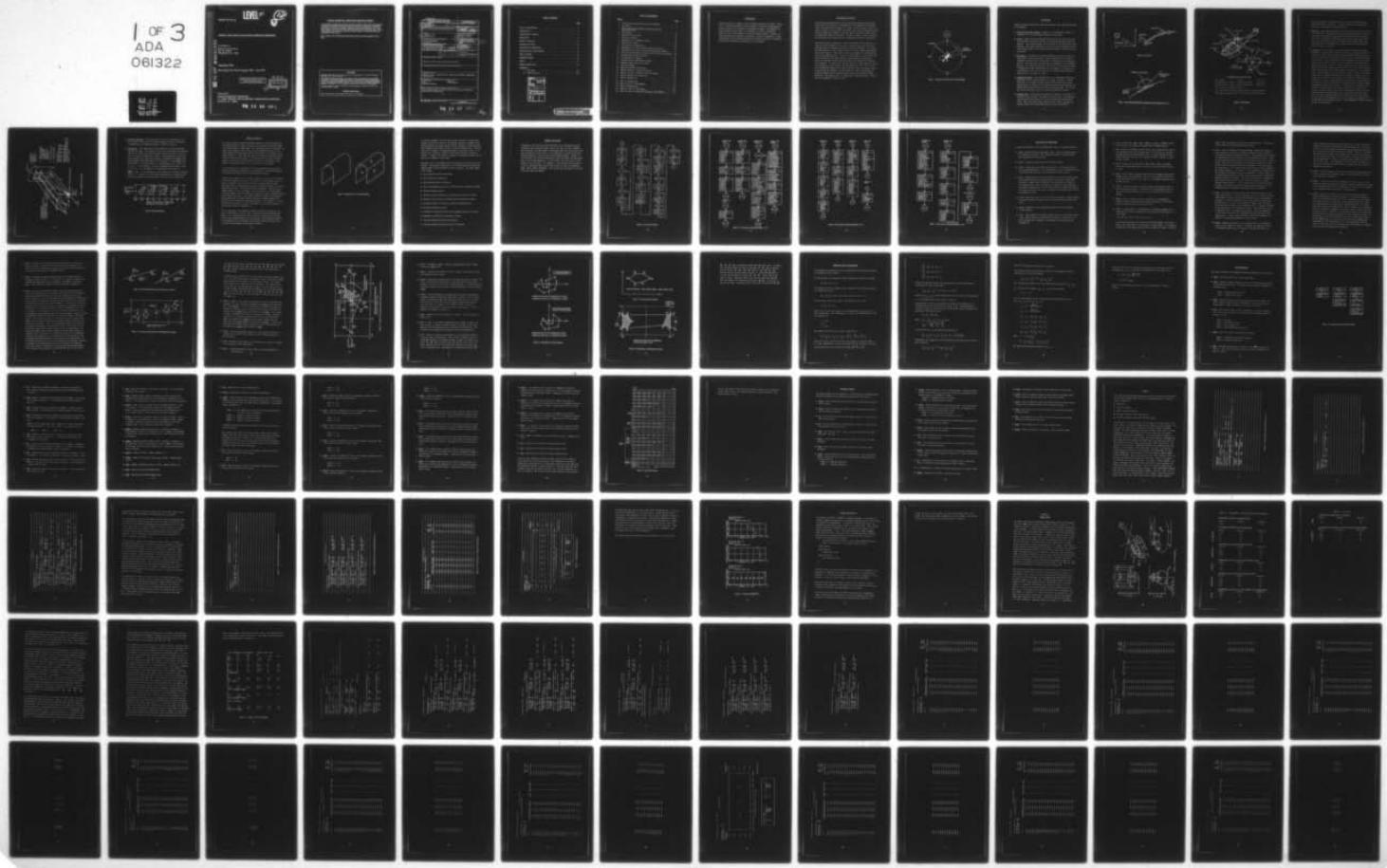
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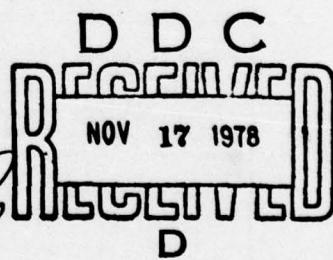
CANOPY SUN GLINT EVALUATION COMPUTER PROGRAM

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September 1978

Final Report for Period August 1976 - July 1977

Approved for public release;
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Prepared for

APPLIED TECHNOLOGY LABORATORY
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Fort Eustis, Va. 23604

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APPLIED TECHNOLOGY LABORATORY POSITION STATEMENT

This document and associated computer program will provide a useful tool in the design and development of helicopter canopies that are to have low sun glint signature characteristics. It will permit the rapid and inexpensive evaluation of various configurations in the initial design phase of a new aircraft as well as retrofit configurations for existing aircraft. Results of this program will advance the technology needed to improve combat effectiveness and survivability of Army aircraft.

Earl C. Gilbert of the Aeronautical Systems Division served as project engineer for this effort.

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INTRODUCTION

Canopy sun glint is a major cue in the visual detection of aircraft. Therefore, it would be desirable to have a method to calculate the magnitude of these sun glints for a proposed or experimental canopy. The computer program described in this report is an attempt to provide an analytical method to solve this problem for smooth reflecting surfaces. However, certain limitations were necessary to maintain a degree of simplicity. These limitations, which include the restriction that any section of a reflecting surface have a quadratic representation, are discussed.

DESCRIPTION OF PROBLEM

An aircraft canopy reflects the rays of the sun at azimuth and elevation angles that are a function of the geometry of the canopy and the orientation of the aircraft with respect to the sun. As the aircraft rotates, as shown in Figure 1, the sun glint also moves; and, as the orientation of the aircraft changes with respect to the sun, more or different canopy panels can become the reflecting surfaces. At any one instant, canopy sun glints may be visible at more than one set of azimuth and elevation coordinates.

The objective then in designing a canopy to minimize visual detection is to limit the size and number of sun glints occurring within the band of coordinates which are probable positions for hostile observers. To evaluate the effectiveness of such a design, there is a choice between experimental testing and analysis. There are at least two problems with experimentally testing new canopy designs. First, if a scaled model is used, getting the proper curvature on the panels could be time-consuming and expensive. Second, if a mockup is used, greater distances are required to simulate a real situation. In either case, the evaluation of a large family of curves, or variations of a basic design, would be difficult. This is not to say that an analytical approach is without problems; there are certain limitations that must be imposed. But the simple fact that a large number of different designs can be evaluated rapidly makes the analytical scheme attractive. The problem then is to develop an analytical method to determine the coordinates of the sun glint.

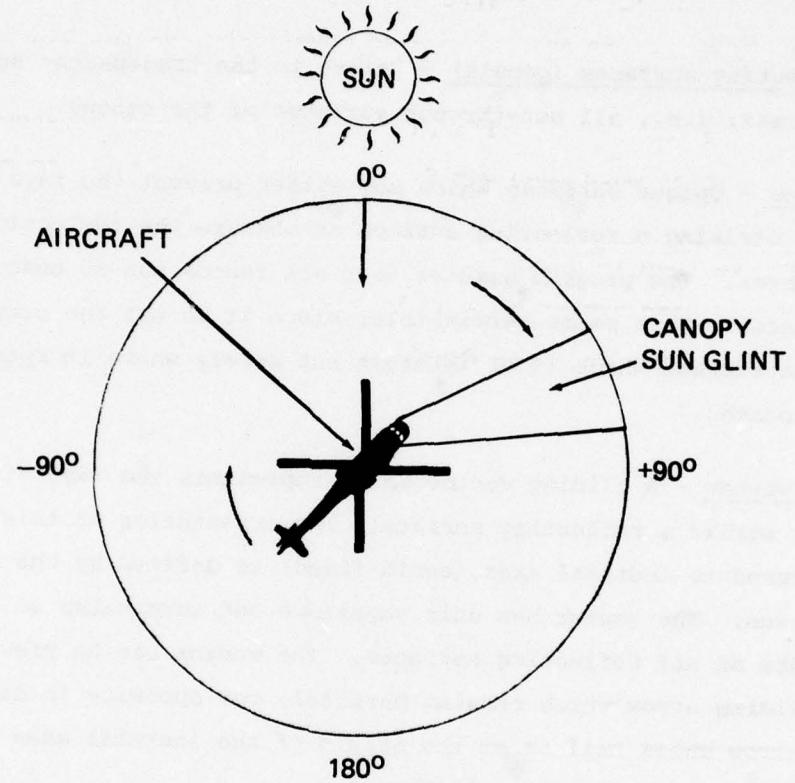


Figure 1. Variation of Sun Glints With Aircraft Heading.

DEFINITIONS

Before proceeding any further with the discussion, some useful definitions are presented:

- Reflecting surfaces (panels) - Refers to the transparent surfaces of interest; i.e., all see-through surfaces of the canopy.
- Fences - Opaque surfaces which may either prevent the rays of the sun from striking a reflecting surface or obscure the reflection from the observer. The program assumes that all fences can be described by flat surfaces. This seems permissible, since it is not the curvature of the fence surface which is of interest but merely where in space the fence is located.
- Sun vector - A sliding vector which represents the rays of the sun as they strike a reflecting surface. The orientation of this vector as referred to inertial axes (earth-fixed) is defined by the elevation of the sun. The vector has unit magnitude and terminates at all boundary points of all reflecting surfaces. The vector can be viewed as simply a sliding arrow which remains parallel, but opposite in direction, to an arrow whose tail is at the origin of the inertial axes and points toward the sun, as shown in Figure 2.
- Reflection vectors - Each boundary point on a reflecting surface has a reflection vector associated with it, as shown in Figure 3. Mathematically, it is formed by taking the sun vector and retaining the portion which is tangent to the reflecting surface while reversing the portion which points toward the surface along the normal. As the normal varies from point to point, so does the reflection vector.
- Inertial axes - Earth-fixed axis system with origin at the center of rotation of the aircraft. The positive X axis points in the direction of the sun while the positive Z axis points away from the center of earth. The Y axis forms right angles with the X and Z axes and has positive sense to the right when facing the sun (see Figure 4).

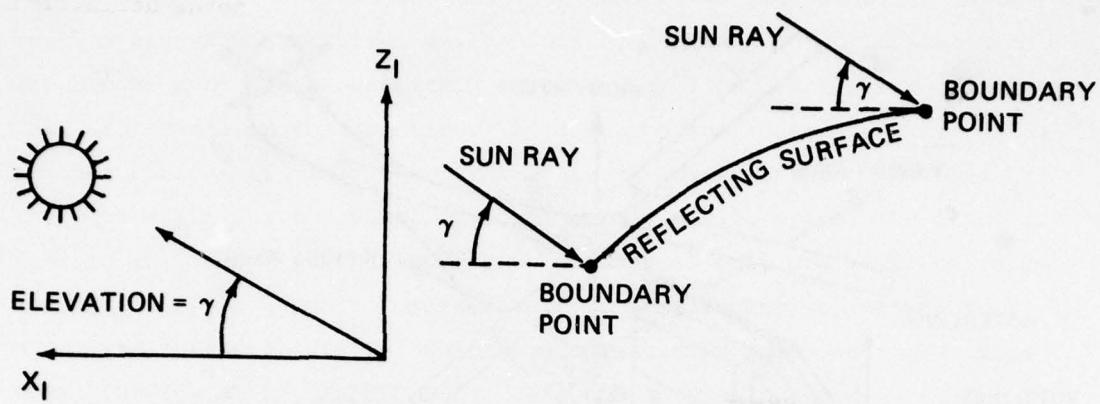


Figure 2. Sun Vector.

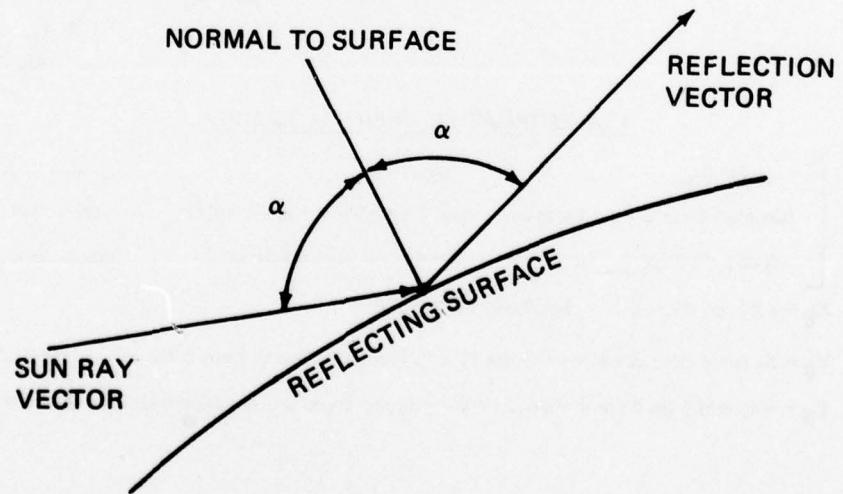
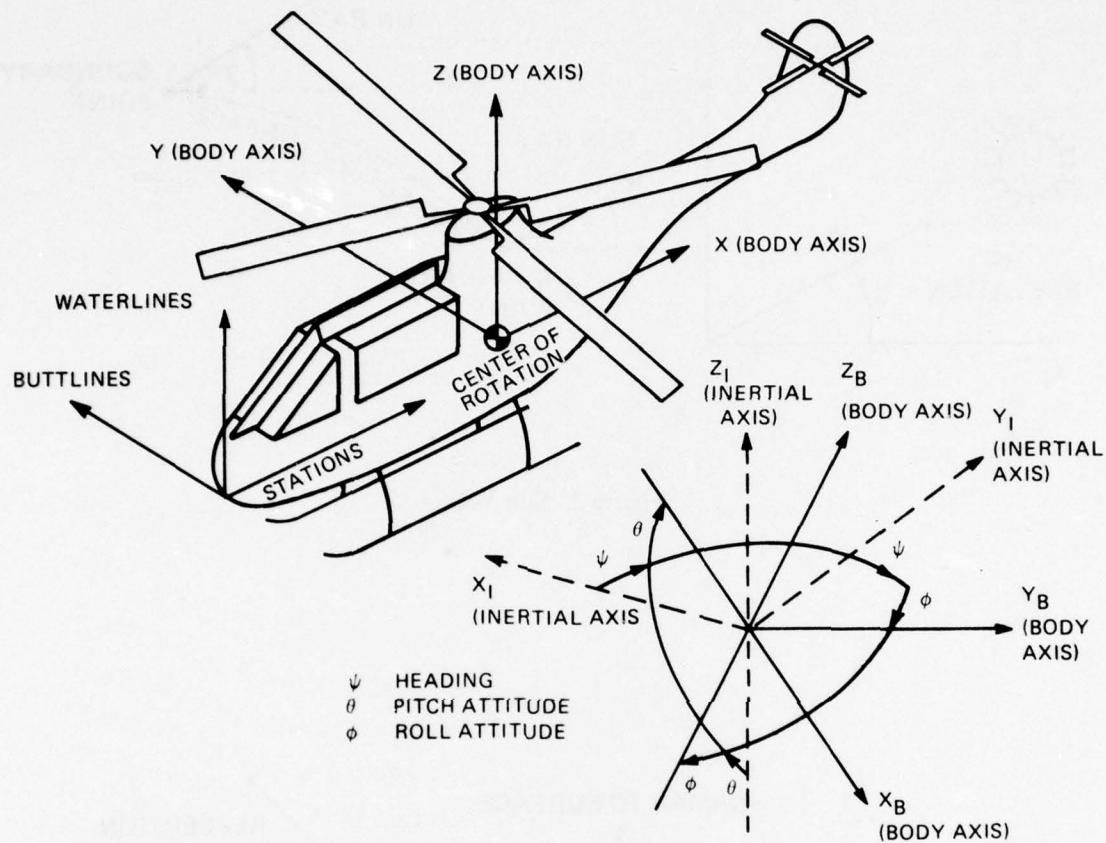


Figure 3. Basic Relationship Between Striking Sun Ray and Reflection Vector.



TRANSFORMATION: INERTIAL TO BODY

$$\begin{bmatrix} -\cos \theta \cos \psi & -\cos \theta \sin \psi & -\sin \theta \\ (\sin \phi \sin \theta \cos \psi - \sin \psi \cos \phi) & (\sin \phi \sin \theta \sin \psi + \cos \psi \cos \phi) & -\sin \phi \cos \theta \\ -(\cos \phi \sin \theta \cos \psi + \sin \psi \sin \phi) & (\sin \phi \cos \psi - \cos \phi \sin \theta \sin \psi) & \cos \phi \cos \theta \end{bmatrix}$$

$$X_B = -X_I \cos \theta \cos \psi - Y_I \cos \theta \sin \psi - Z_I \sin \theta$$

$$Y_B = X_I (\sin \phi \sin \theta \cos \psi - \sin \psi \cos \phi) + Y_I (\sin \phi \sin \theta \sin \psi + \cos \psi \cos \phi) - Z_I \sin \phi \cos \theta$$

$$Z_B = -X_I (\cos \phi \sin \theta \cos \psi + \sin \psi \sin \phi) + Y_I (\sin \phi \cos \psi - \cos \phi \sin \theta \sin \psi) + Z_I \cos \phi \cos \theta$$

Figure 4. Axis Systems.

Aircraft headings are measured relative to this axis system and are positive as shown in Figure 1. Sun glint position is also measured relative to these axes as shown in Figure 5.

- Body axes - Body-fixed axis system with origin at the center of rotation of the aircraft. The positive X axis points toward the tail of the aircraft while the positive Z axis points toward the top of the aircraft. The Y axis is positive out the right side of the aircraft and, along with the X and Z axes, forms an orthogonal system. These axes are assumed parallel to the station line, buttline, and waterline reference system as shown in Figure 4.
- Sun glints - In this report, a sun glint refers to the single image produced by one reflective surface for one set of aircraft and sun positions. If the reflection vectors are extended, as shown in Figure 5, until they intersect an imaginary cylinder in space, the image produced is a sun glint. The whole thing can be visualized as if the aircraft were sitting in the center of a circular stadium and the stands were filled with hostile observers. These stands are strictly vertical and form the rim of the stadium. The reflection vector from each boundary point, when extended, will strike someone in the stands. If each person that is struck stands up, the outline of an image is formed. The image formed from the reflection vectors of one reflective surface will be called the sun glint of that surface at the specified aircraft and sun conditions. Since the radius (distance to observers) of the stadium is a fixed input value, the location of any person in the stands is defined completely by specifying two angles. The two angles are measured relative to the inertial axes. To locate a person, first rotate the X inertial axis about the Z axis until it intersects the stands directly below the person (azimuth angle); then rotate the X axis about the Y axis until the X axis points directly at the person (elevation angle). In the same way the boundary points of the sun glint are defined by an azimuth angle and an elevation angle. These coordinates (angles) form the results of this program.

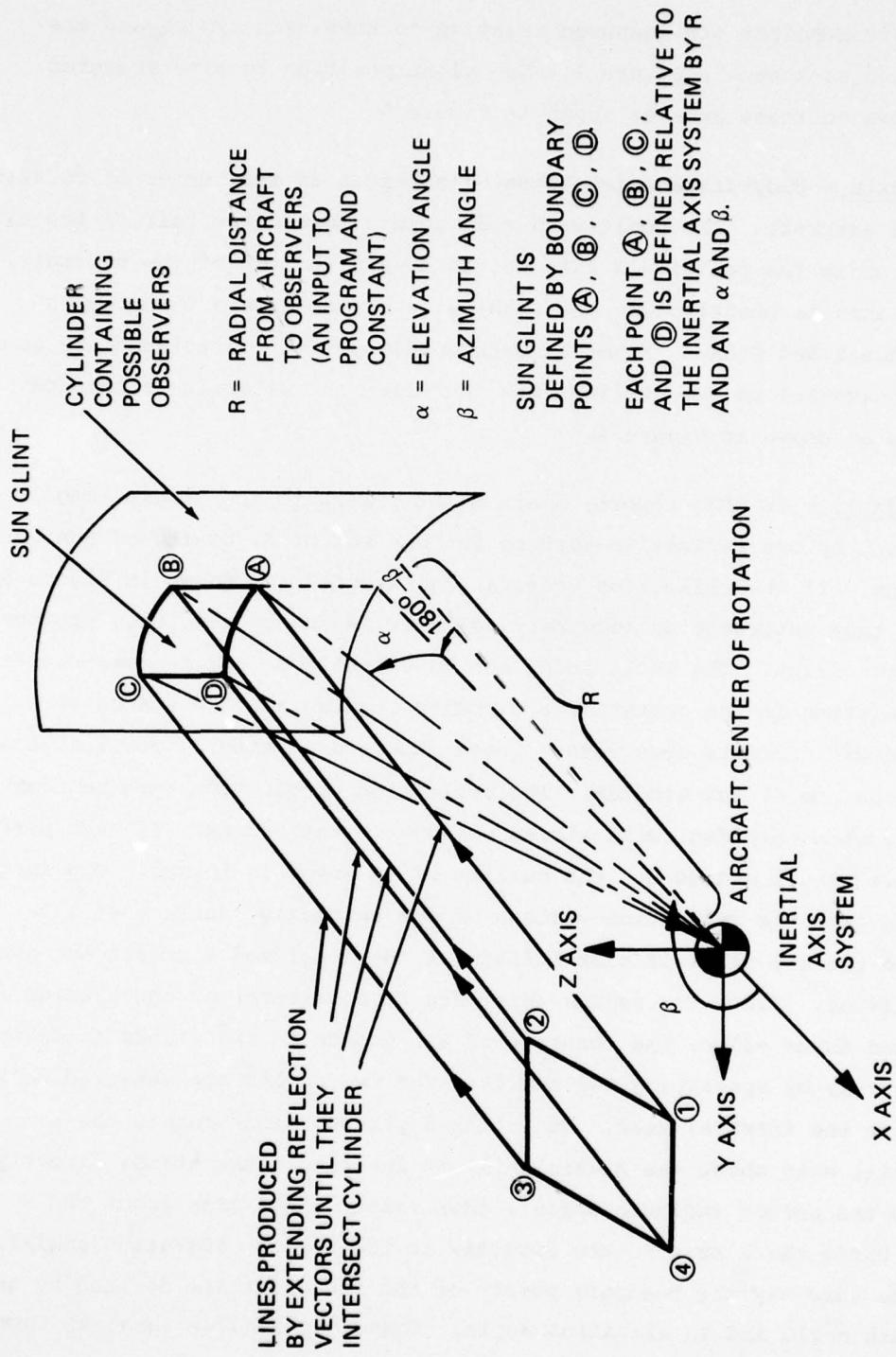


Figure 5. Definition of Sun Glint.

- Sun glint signature - This is the plot of all the coordinates of all the sun glints from all the reflective surfaces for one sun elevation. A sample plot for an imaginary canopy is shown in Figure 6.
- Probability - The probability of being observed is defined to be the ratio of two areas. The numerator is the area within the glint signature plot which is swept out by the reflected canopy sun glints as the aircraft is rotated 360 degrees in heading. The denominator is the total area enclosed by the rectangle formed by the maximum and minimum allowable elevation and azimuth angles of the observer. As an example, consider Figure 6. The area swept out is approximately $(135 - (-150)) \times (5 - (-5)) = 2,850$ degrees². The total allowable area is $(180 - (-180)) \times (10 - (-10)) = 7,200$ degrees². The probability would then be $\frac{2,850}{7,200} = 0.396$. In the calculation of the swept area, overlap is ignored; that is, if a portion of the sun glint signature plot is covered by the sun glints from more than one surface, the area is only counted once.

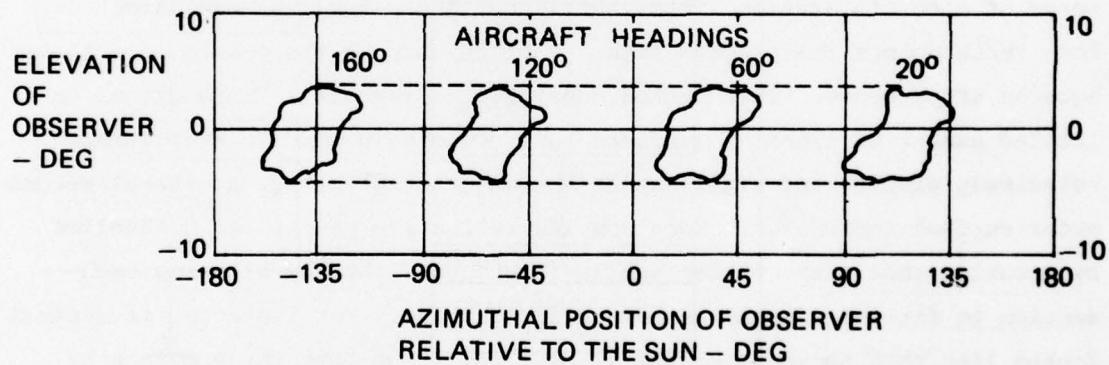


Figure 6. Sun Glint Signature.

METHOD OF SOLUTION

The rays of the sun, treated as parallel vectors, strike the canopy and are reflected by the canopy. The method of solution is basically simple; the trimmings make it complicated. The reflecting surface is assumed to be smooth, and therefore, the reflection vector becomes a function of the angle between the striking sun rays and the normal to the surface, as was shown in Figure 3. This angle may vary as the relative position between the aircraft and the sun varies, for instance, due to aircraft heading changes. It could also vary from point to point on the surface as the normal to the surface varies, as in the case of a curved panel.

It is necessary then to determine at all the boundary points of all the reflective surfaces the normals to those surfaces. One method considered was to input the normals for all the reflective surfaces.

However, these normals are not readily available in most cases. Therefore, an alternate approach was taken. The location of the boundary points in terms of aircraft station lines, buttlines, and waterlines was used. From these points the program fits a surface through the points in a least-squares error sense. This method does have a drawback. There are an unlimited number of types of surfaces that could be used. To keep things relatively simple, and still handle the majority of cases, a general second-order surface was chosen. More complex reflective panels can be handled by breaking them into smaller panels (sections), thereby allowing each section to fit a different second-order equation. For instance, if a panel looked like that shown in Figure 7, breaking it up into three panels as shown should yield an acceptable fit.

Once the equation of the reflective surface is known, the normal at any point on the surface is known. Then, selecting an aircraft orientation (pitch, roll, and heading) and sun elevation, the angle between the sun vector and any surface normal can be calculated. The reflection vector at each boundary point is then known; and from the reflection vectors, the sun glints can be formed, as was shown in Figure 5.

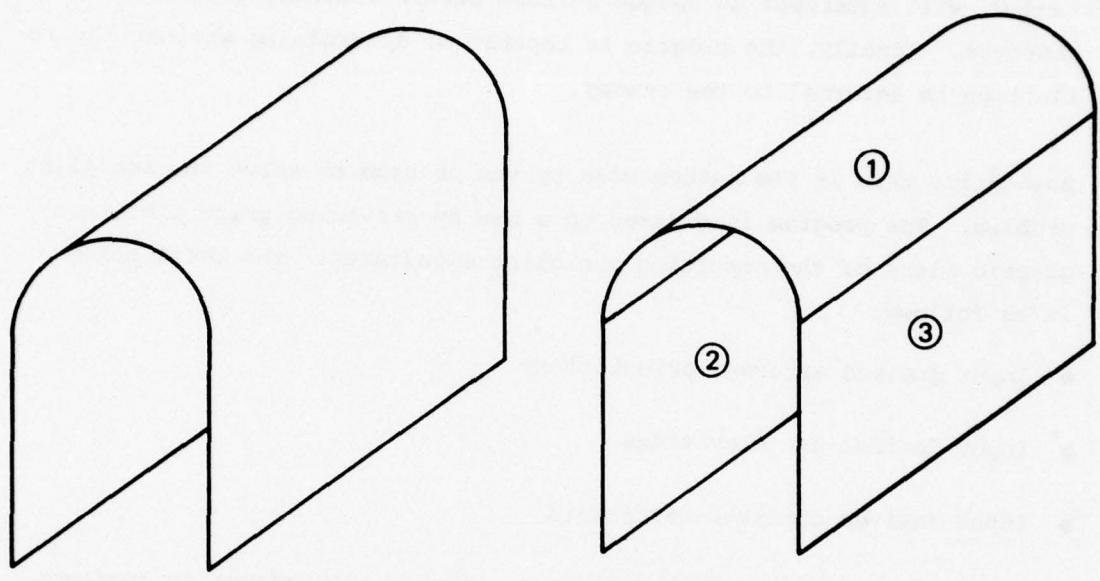


Figure 7. Segmentation of a Compound Surface.

It has been assumed so far that the sun rays are free to strike every boundary point on the surface. The program, however, is capable of determining whether an opaque surface (fence), defined by boundary points, comes between the sun and a reflective surface boundary point. The program is also capable of determining whether a reflection vector, if extended, will intersect an opaque surface before reaching a possible observer. Finally, the program is capable of determining whether the reflection is internal to the canopy.

Basically, this is the method used by the program to solve the sun glint problem. The program is dressed up a bit by providing print plots and graphic plots of the resulting sun glint signatures. The total scheme is as follows:

- Input desired aircraft orientations
- Input desired sun elevations
- Input desired observer restraints
- Input the boundary points for all fences and all reflective surfaces
- Curve-fit boundary points
- Calculate relative position of sun vector and reflective surface
- Determine if any fences lie between sun and reflective surface
- Calculate normals to reflective surface at boundary points
- Calculate reflection vectors
- Determine if reflection vectors when extended intersect any fences
- Determine if reflection is internal to canopy
- Calculate observer positions (sun glints)
- Calculate probabilities from sun glint signatures.

PROGRAM FLOW CHARTS

Presented in this section are the flow charts for the executive (main) routine and each subroutine which contains at least one CALL statement. The blocks that are shown within a particular flow chart refer only to the CALL statements in that routine. If a subroutine is called which calls another subroutine, only the first CALL statement appears. However, in the flow chart for the first subroutine called will appear the CALL statement to the second subroutine. In other words, only one level of calls appear in any flow chart. For example, in the executive routine there is a call to subroutine CFITF, which would result in calls to subroutines NORM, CURFIT, SIGNF, AND EQNSOL. This can be seen by looking at the flow charts for CFITF and CURFIT.

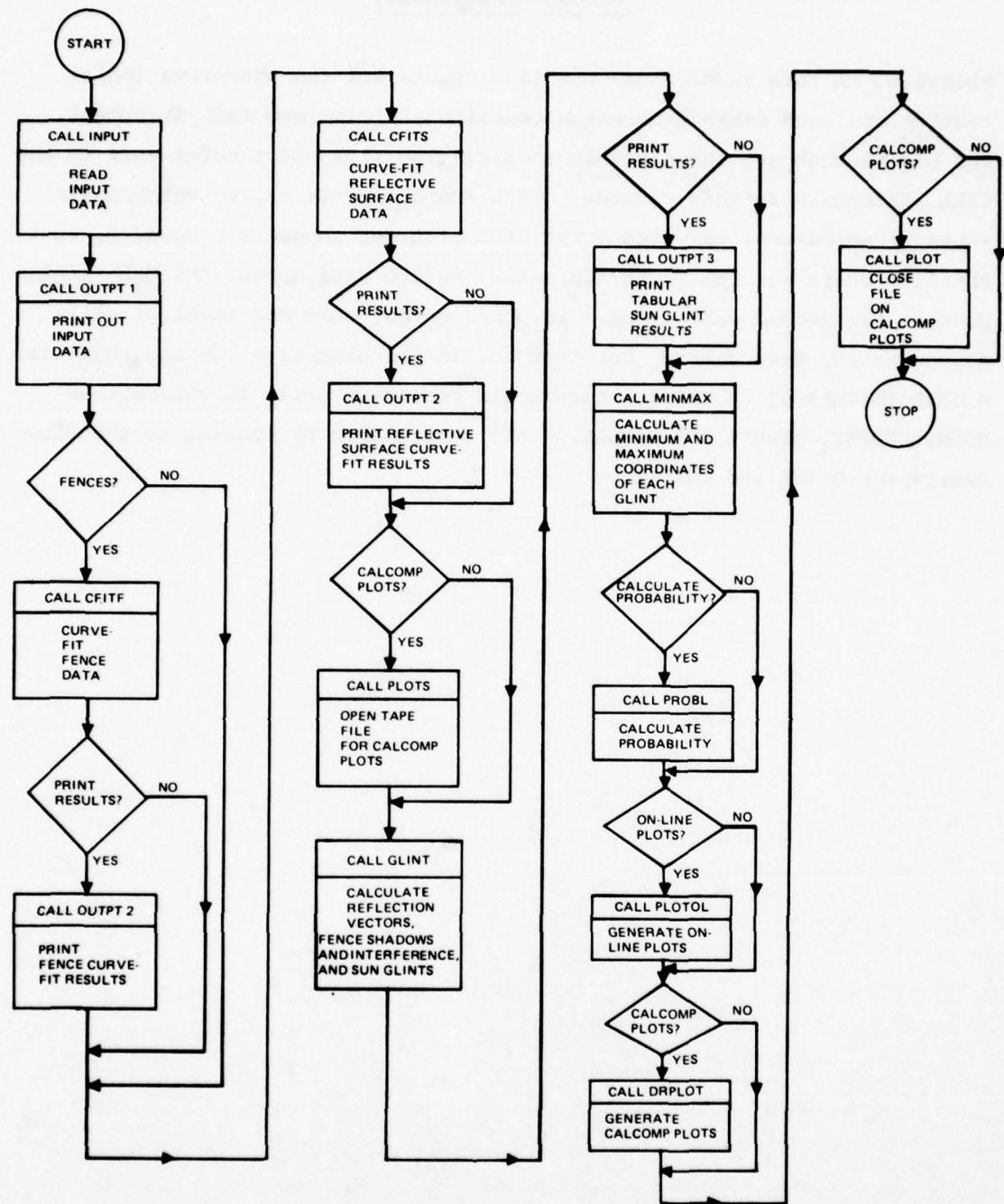


Figure 8. Executive Flow Chart

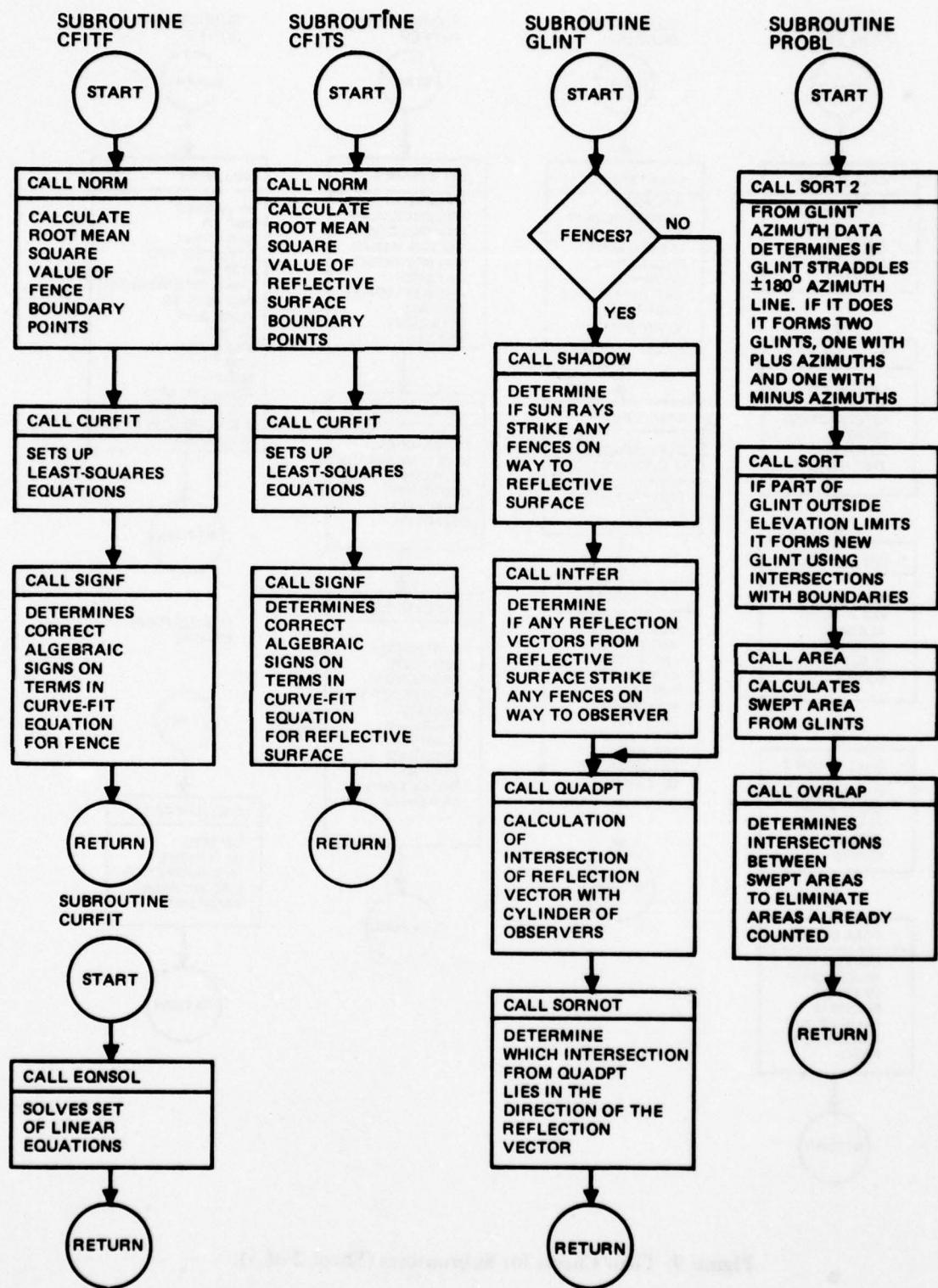


Figure 9. Flow Charts for Subroutines (Sheet 1 of 3).

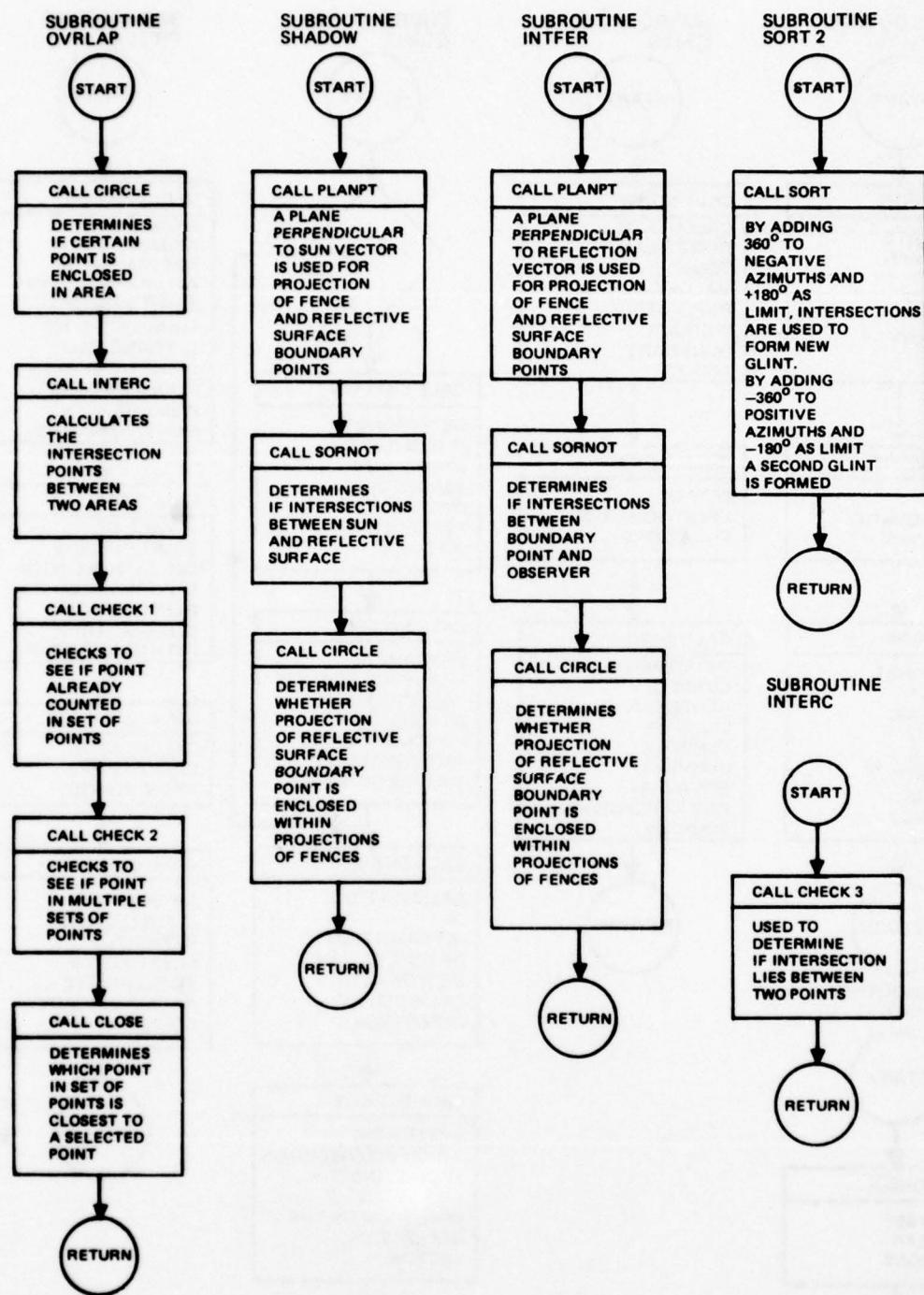


Figure 9. Flow Charts for Subroutines (Sheet 2 of 3).

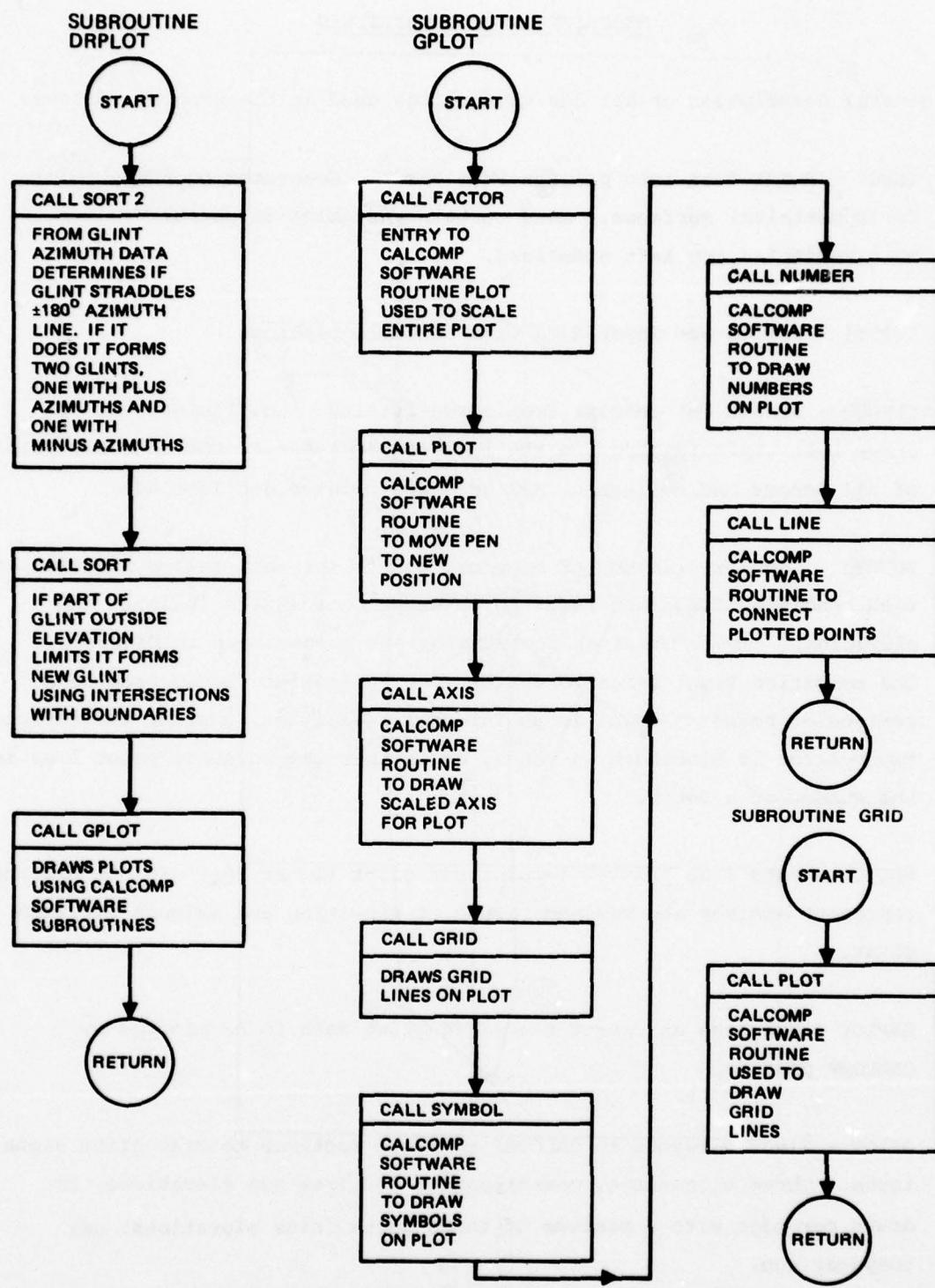


Figure 9. Flow Charts for Subroutines (Sheet 3 of 3).

DESCRIPTION OF SUBROUTINES

A general description of all the subroutines used in the program follows:

- **INPUT** - Reads data into program from cards. Generates boundary points for symmetrical surfaces. Sets certain variables to default values, when variables are left undefined.
- **OUTPT1** - Prints out input data with suitable headings.
- **OUTPT2** - Prints out results from curve-fitting. Coefficients of the curve best approximating, in the least-squares sense, the boundary points of all fences and reflective surfaces are printed and labeled.
- **OUTPT3** - Tabular printout of boundary points for each reflective surface with condition flags and required observer coordinates (azimuth and elevation). These observer coordinates are those shown in Figure 5. The condition flags refer to whether the reflection vector from each particular boundary point is an internal reflection, whether the reflection vector is blocked by a fence, or whether the boundary point lies in the shadow of a fence.
- **PLOTOL** - Uses line printer to plot sun glint signatures. Points plotted represent minimum and maximum values of elevation and azimuth for each glint.
- **DRPLOT** - Prepares and sorts tabulated glint data to be plotted by CALCOMP plotter.
- **GPLOT** - Gives commands to CALCOMP software routines to draw glint signatures. Three signatures, corresponding to three sun elevations, are drawn per page with a maximum of three pages (nine elevations) per computer run.

- PLOTS, FACTOR, PLOT, AXIS, SYMBOL, NUMBER, and LINE - CALCOMP software functions which are supplied by CALCOMP BASIC SOFTWARE package.
- GRID - Normally a CALCOMP software routine; this subroutine was written and included as part of the program since it is not part of the basic packages supplied to some CALCOMP users. If system already has GRID, modify subroutine name and call statement to make it unique, i.e., GRIB instead of GRID.
- CFITF - Along with the subroutines NORM, CURFIT, SIGNF, and EQNSOL, curve-fits the boundary points of a fence to a plane (linear) surface in space.
- NORM - In an attempt to minimize numerical errors during curve-fitting, boundary points are normalized. This routine calculates a norm which is the root-mean-square value of the set of coordinates representing the boundary points.
- CURFIT - Sets up the normal coefficient matrix and constant terms for the least-squares solution. Calculates the root-mean-square error of the fit by plugging boundary points into the equation of the fitted surface.
- EQNSOL - Solves a system of linear, homogeneous or inhomogeneous, equations using the technique of Gaussian elimination. It is used to solve for coefficients of the fitted surface.
- SIGNF - A surface which is represented by an equation in terms of its coordinates can also be represented by the negative of the equation. In other words,

$$Ax + By + Cz + D = 0 = -Ax - By - Cz - D.$$

In one case, the program will generate an outward normal to the canopy while in the other case it will generate an inward normal. In subsequent calculations, the program requires that the normal be outward. Therefore,

SIGNF is used to determine correct sign on coefficients. It does this using tests on the two possible normal vectors.

- CFITS - A set of boundary points representing a reflective surface is used to produce, in general, a quadratic equation (a plane surface is considered a degenerate quadratic surface). This equation is an analytical representation of the points in a least-squares error sense. There are three options available when trying to match an equation to a set of points. The first is to specify the coefficients, either all or some. In this case the calculated values are overridden by the input values. The second option is to specify which coefficients are to be included in the matching equation. For instance, if the program user wished the surface through the boundary points to be quadratic in x while linear in z , he would specify through his inputs that the other terms in the general quadratic equation be ignored. The third option, which is probably the most useful, starts out by trying to match the points to a linear equation (plane surface). If this equation fails to satisfy the error criteria, the program continues to form quadratic combinations. The combination with the minimum least-squares error is selected.
- GLINT - Calculates the sun glints for each reflective surface as aircraft varies heading. First the components of the sun vector are transformed from the inertial axes to the body axes by the transformation in Figure 4. Then each boundary point of the reflective surfaces is checked to see if it lies in the shadow of a fence. Next the normal to the reflective surface at each boundary point is calculated followed by the calculation of the reflection vectors. Each reflection vector is checked to see if it intersects any fences. Then the components of the reflection vectors are transformed from the body axes to the inertial axes. Finally, these reflection vectors are extended as in Figure 5 to find the coordinates of each sun glint.
- SHADOW - Answers the question, "Is a boundary point of a reflective surface in the shadow of a fence?" To answer this question, first it is determined whether any of the planes containing the fences are

located between the boundary point and the sun. If a fence plane lies between the sun and the reflective surface boundary point, the intersection of the line formed by projecting the boundary point along the sun vector with the plane of the fence is saved. Finally, it is determined whether any of the saved intersections lie within the boundaries of their respective fences. If any of them do, then the reflective surface boundary point is indeed in a shadow and the ISHAD flag is set to one for this point. The coordinates of the sun glint for this point are still calculated and tabulated but not plotted and not used in the probability calculation.

- PLANPT - Given the equation of a plane surface and the representation of a line by a point and direction cosines, the intersection of the line with the plane is calculated.
- SORNOT - Given a point and the coordinates of a unit vector, a line through the point with the same direction cosines as the vector is formed. It is then determined whether a second point on the line is in the positive or negative sense of the vector (see Figure 10).
- INTFER - Answers the question, "Does a reflection vector intersect a fence before reaching a possible observer?" To answer this question, first it is determined whether any of the planes containing the fences are located between the reflective surface boundary point from which the reflection vector emanates and any possible observer. If a fence plane lies between the boundary point and the observer, the intersection of the line formed by extending the reflection vector with the plane of the fence is saved. Finally, it is determined whether any of the saved intersections lie within the boundaries of their respective fences. If any of them do, then the reflection vector does indeed strike a fence and the INTFER flag is set to one for the reflective boundary point corresponding to the reflection vector. The coordinates of the sun glint for this point are still calculated and tabulated but not plotted and not used in the probability calculation.

- QUADPT - Calculates the intersection or intersections of a line with a quadratic surface. Inputs to the subroutine are the coordinates of a point on the line, the direction cosines of the line, and the coefficients of the equation describing the quadratic surface.
- MINMAX - Determines minimum and maximum values of the elevations and azimuths for each sun glint. At this point, 0.25 degree is added to the maximums and subtracted from the minimums to represent sun dispersion angle. These values are plotted and used to determine probability.
- PROBL - As the aircraft changes heading, with fixed sun elevation and fixed aircraft pitch and roll attitudes, the sun glint from a reflective surface may change position on the sun glint signature plot, as was shown in Figure 6. If all intermediate positions between the discrete sun glint positions calculated are assumed possible, then there is a continuous area swept out during the heading changes. The total of all the areas swept by all the reflective surfaces, but counting any overlap only once, divided by the total area possible will yield the probability for this sun elevation. In programming a method to calculate the swept area, certain assumptions and simplifications were found necessary. Figure 11 shows a hypothetical example of the motion of a sun glint from one reflective surface. The four points shown for each sun glint are the results from MINMAX, which represent the minimum and maximum observer elevations and azimuths for this glint. This is simplification number one, to use only the minimum and maximums to represent each glint. This allows for the generation of the polygon shown in Figure 11 by the dotted lines. The single point (C) at zero aircraft heading is another simplification or assumption and is caused when all the boundary points of the reflective surface at this heading are in the shadow of one or more fences or when the generated reflection vectors are blocked by one or more fences. In other words, there is no glint for this surface at zero heading. However, at headings of ± 80 degrees there are glints; therefore, the approach chosen was to place a single point at the average azimuth and elevation of the observers had the fences not been there and thus a glint created.

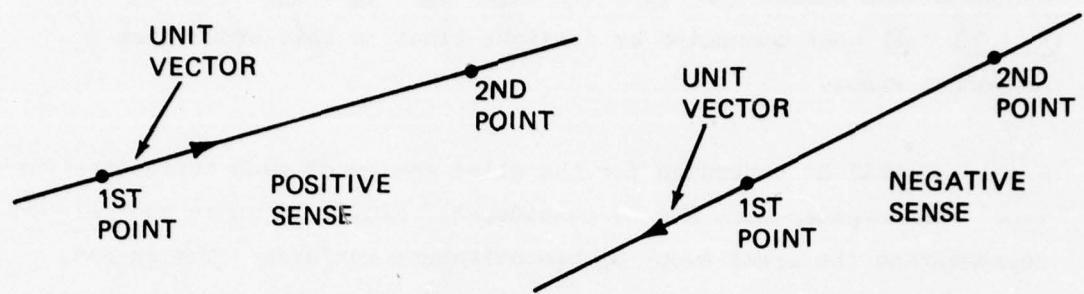


Figure 10. Determination of Positive Sense Between Two Points.

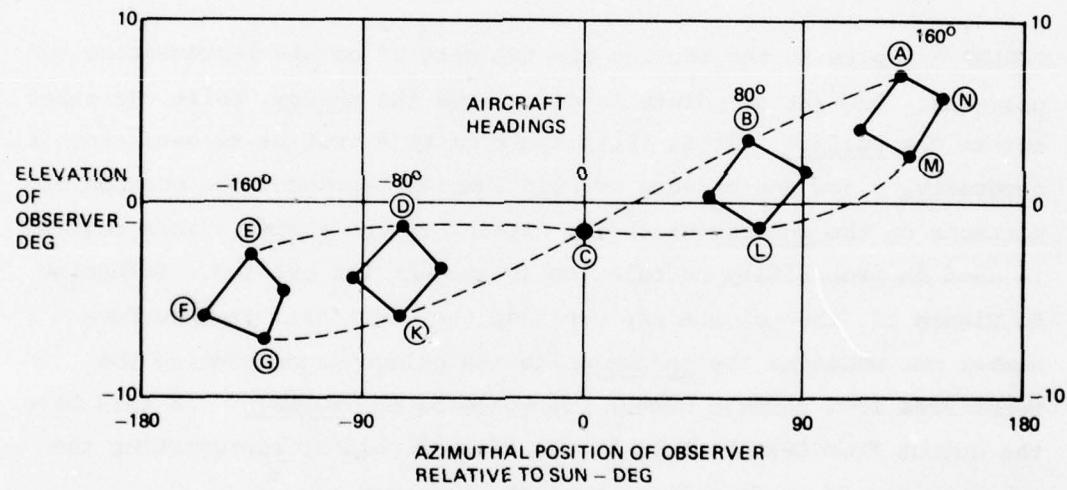


Figure 11. Area Swept by Sun Glints From Single Reflective Surface.

By doing this, the area is tapered to zero on both sides of the point C. As shown, the points F E D C B A N M L C K G F when connected by straight lines in this order form a polygonal shape.

A polygon will be generated for the glint motion of each reflective surface. The overlap must now be considered. Figure 12 shows two polygons representing the areas swept by two different surfaces. The shaded area represents the common area to both polygons. The program treats polygon ① as if polygon ② does not exist; thus the shaded area is included in ①. Then polygon ② is broken up into two polygons formed by the points P Q R X Y and S T U V W T. When finished there are three polygons, instead of two, but with no common area.

- OVRLAP - Inputs to the routine are two sets of points representing two polygons. One set of points is designated the shadow, while the other set is the surface. It is the purpose of this routine to generate, if necessary, a new set or sets of points which represent the portion or portions of the surface which lie outside of the shadow. This routine is used in probability calculation to account for overlap. Referring to Figure 12, the polygon representing the swept area from surface number one would be the shadow while the polygon representing the swept area from surface number two would be the surface. In this case, the output from OVRLAP would be two sets of points, representing the two portions of surface lying outside of shadow.
- INTERC - Finds intersections, if any, of a line between two points in a plane with the set of lines given by a set of connected points in the same plane.
- CLOSE - Determines which point in a set of points in a plane is closest to a given point in the same plane.
- CHECK 1 - Determines whether a given point is already contained in a set of points.

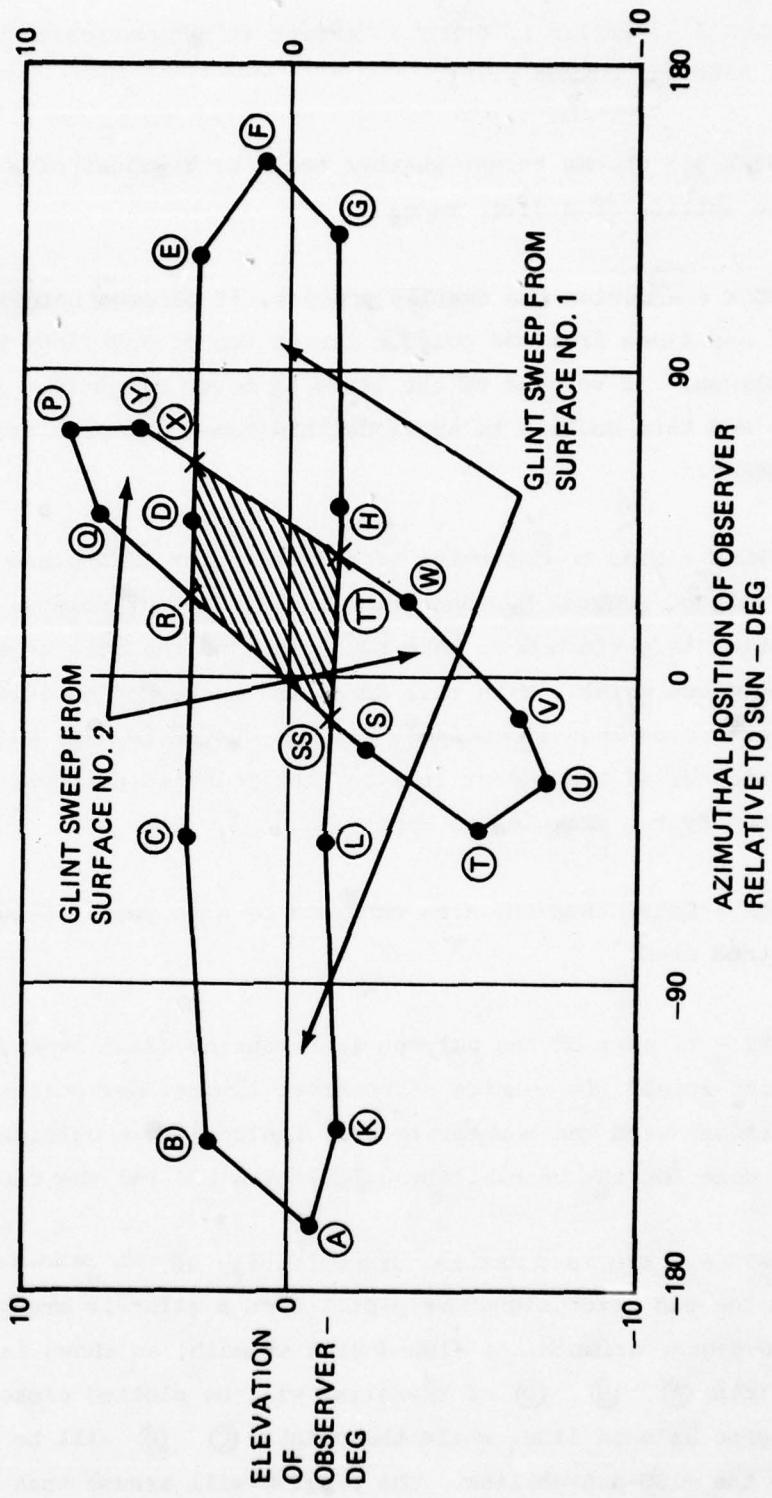
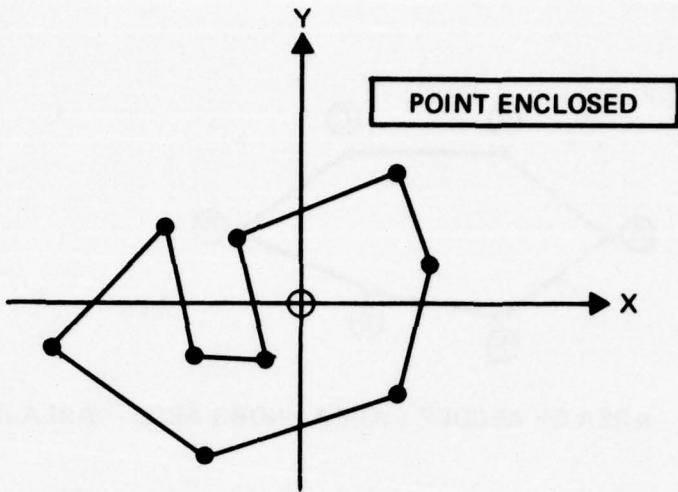
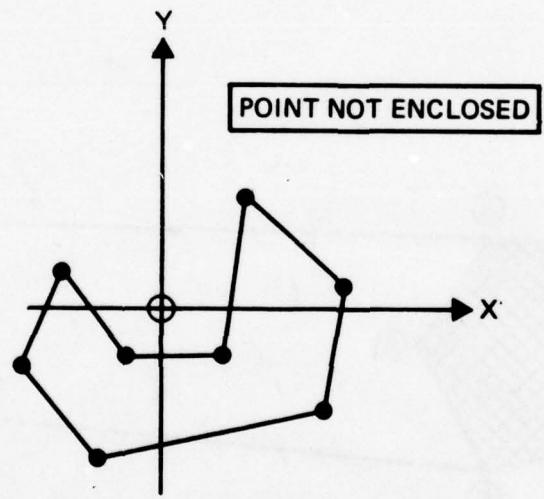


Figure 12. Overlap of Areas Swept by Glints From Two Reflective Surfaces.

- CHECK 2 - Similar to CHECK 1, except it automatically checks a number of sets for common point.
- CHECK 3 - Checks to see whether the X or Y values of a point in a plane are outside of a given range.
- CHECK 4 - During the overlap process, it becomes necessary to check to see if any lines from one polygon lie on top of any lines from the other polygon. If so, one of the lines is moved slightly. It became necessary to add this routine to avoid certain numerical problems in the subroutine OVRLAP.
- CIRCLE - Used to determine if a given point in a plane is enclosed by a polygon created by connecting a given set of points. First, a translation is performed to make the origin of the axis system lie on top of the given point. With this done, the number of positive or negative X-axis crossings is counted. If the number is odd, the point is enclosed; if the number is even, the point is not enclosed. This is shown by two examples in Figure 13.
- AREA - Determines the area enclosed by a polygon. Figure 14 shows the method used.
- SORT - If part of the polygon representing glint sweep, or part of the glint itself, is outside of observer limits, new points at the intersections with the boundaries will replace those outside limits. This is done for the probability calculation and for the CALCOMP plots.
- SORT2 - There is a natural discontinuity at the ± 180 -degree azimuths on the sun glint signature plot. When a glint is wrapped around the 180-degree azimuth, or -180-degree azimuth, as shown in Figure 15, the points **(A)** **(B)** **(E)** of the glint will be plotted close to the -180-degree azimuth line, while the points **(C)** **(D)** will be plotted close to the +180-degree line. The program will assume that the glint **(A)**



NUMBER OF POSITIVE X CROSSINGS IS 1 (ODD)
NUMBER OF NEGATIVE X CROSSINGS IS 3 (ODD)



NUMBER OF POSITIVE X CROSSINGS 2 (EVEN)
NUMBER OF NEGATIVE X CROSSINGS 2 (EVEN)

Figure 13. Determination of Point Enclosure.

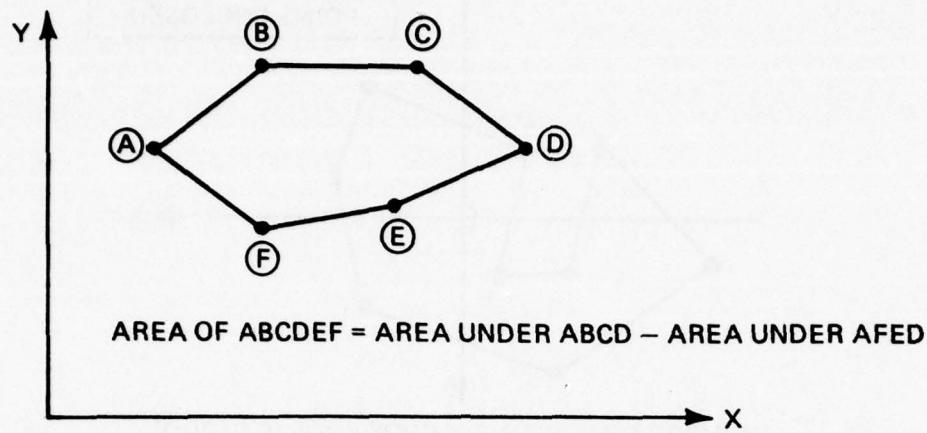


Figure 14. Area Enclosed by Polygon.

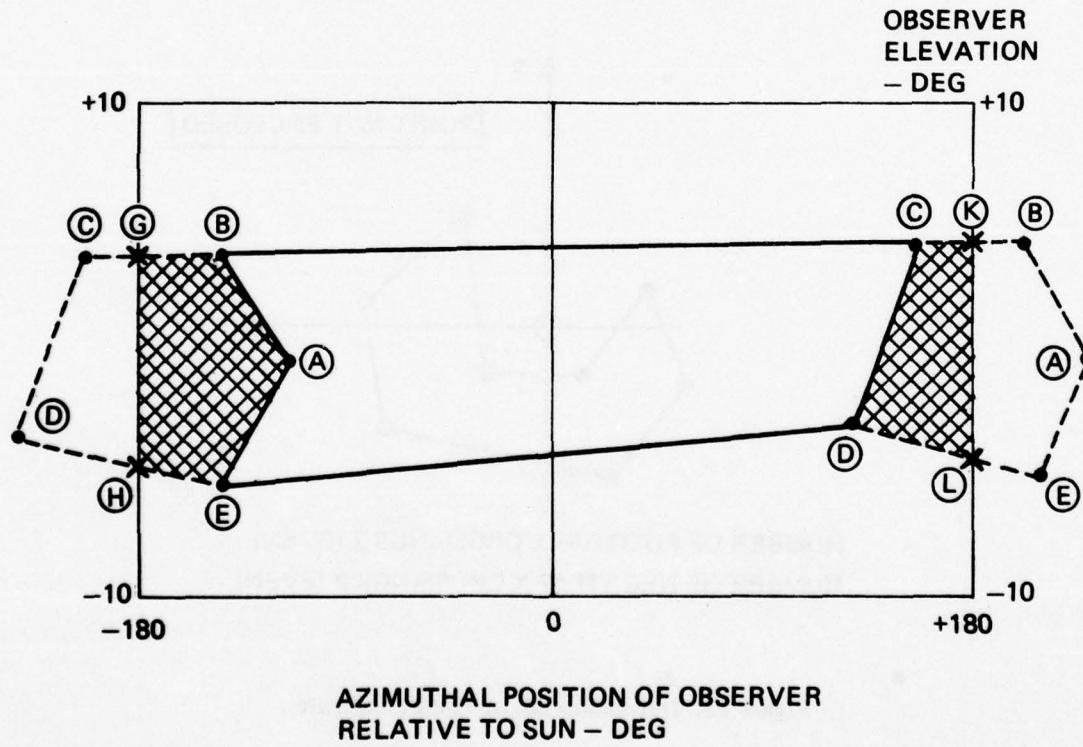


Figure 15. Discontinuity at ± 180 Degrees Azimuth.

(B) (C) (D) (E) is connected as shown with the solid lines. To handle this situation properly, the glint (A) (B) (C) (D) (E) is replaced with two glints, (A) (B) (G) (H) (E) and (C) (K) (L) (D). The new glints are shown shaded in Figure 15. The method employed is to first add 360 degrees to points (A) (B) (E), moving them to the right as shown. Subroutine SORT is then used to find points (K) (L). Next, 360 degrees are subtracted from points (C) (D), moving them to the left as shown. Subroutine SORT is then used to find points (G) (H).

EQUATIONS USED IN THE PROGRAM

The mathematical expressions used in calculating the sun glint coordinates are presented in this section.

The general form of the equation used to represent all fences is a plane,

$$Ax + By + Cz + D = 0.$$

The general form of the equation used to represent all reflective surfaces is a quadratic surface,

$$Ax^2 + By^2 + Cz^2 + Dx + Ex + Fy + Gy + Hx + Kz + L = 0.$$

The sun vector referred to inertial coordinates is of the form

$$\bar{s} = ai + bj + ck,$$

where (\bar{i} , \bar{j} , \bar{k}) are unit vectors in the direction of the inertial x , y , z axes, respectively. The coefficients a , b , and c are defined using γ , the sun elevation, as

$$a = -\cos \gamma$$

$$b = 0$$

$$c = -\sin \gamma.$$

The normal to the reflective surface in body axes is

$$\bar{N}_B = N_{B_x} \bar{i}_B + N_{B_y} \bar{j}_B + N_{B_z} \bar{k}_B = \frac{\partial \phi}{\partial x} \bar{i}_B + \frac{\partial \phi}{\partial y} \bar{j}_B + \frac{\partial \phi}{\partial z} \bar{k}_B,$$

where (\bar{i}_B , \bar{j}_B , \bar{k}_B) are unit vectors in the direction of aircraft body x , y , z axes, respectively, and the coefficients $\frac{\partial \phi}{\partial x}$, $\frac{\partial \phi}{\partial y}$, and $\frac{\partial \phi}{\partial z}$ are the first partial derivatives of the equation representing the surface.

$$\frac{\partial \phi}{\partial x} = 2Ax + Dy + Ez + G$$

$$\frac{\partial \phi}{\partial y} = 2By + Dx + Fz + H$$

$$\frac{\partial \phi}{\partial z} = 2Cz + Ex + Fy + K.$$

Before the reflection vector can be calculated, the sun vector must be represented in terms of body axes.

$$(S_{x_B}, S_{y_B}, S_{z_B})^T = D (-\cos \gamma, 0, -\sin \gamma)^T$$

where $(S_{x_B}, S_{y_B}, S_{z_B})$ are the components of the sun vector in body axes and

D is the transformation matrix shown in Figure 4.

The magnitude of the sun vector along the normal to the surface is the projection of the sun vector onto the normal vector. Mathematically, it is the dot product of the sun vector and the normal vector, divided by the magnitude of the normal vector.

$$S_N = \bar{S}_B \cdot \bar{N}_B / |\bar{N}_B|,$$

where $\bar{S}_B = S_{x_B} \bar{i}_B + S_{y_B} \bar{j}_B + S_{z_B} \bar{k}_B$ and

$$|\bar{N}_B| = \sqrt{N_{x_B}^2 + N_{y_B}^2 + N_{z_B}^2}.$$

Using the value of S_N , the reflection vector \bar{R}_B is

$$\bar{R}_B = R_{x_B} \bar{i}_B + R_{y_B} \bar{j}_B + R_{z_B} \bar{k}_B = \bar{S}_B - (2 S_N) \bar{N}_B / |\bar{N}_B|.$$

Transforming the components of the reflection vector from body to inertial axes is done by

$$(R_{x_I}, R_{y_I}, R_{z_I})^T = D^T (R_{x_B}, R_{y_B}, R_{z_B})^T,$$

where D^T is transpose of the matrix of Figure 4.

Each boundary point of the reflective surface is transformed from body coordinates to inertial coordinates.

$$(x_{S_I}, y_{S_I}, z_{S_I})^T = D^T (x_{S_B}, y_{S_B}, z_{S_B})^T$$

The intersections caused by the line passing through the point $(x_{S_I}, y_{S_I},$

$z_{S_I})$ with direction numbers (R_x, R_y, R_z) and the cylinder

$$x^2 + y^2 = R^2,$$

which is the imaginary surface containing the possible observers, are

$(x_{0_1}, y_{0_1}, z_{0_1})$ and $(x_{0_2}, y_{0_2}, z_{0_2})$.

$$x_{0_1} = (-b + \sqrt{b^2 - 4ac})/2a$$

$$x_{0_2} = (-b - \sqrt{b^2 - 4ac})/2a$$

$$y_{0_1} = y_{S_I} + (R_y/R_x) (x_{0_1} - x_{S_I})$$

$$y_{0_2} = y_{S_I} + (R_y/R_x) (x_{0_2} - x_{S_I})$$

$$z_{0_1} = z_{S_I} + (R_z/R_x) (x_{0_1} - x_{S_I})$$

$$z_{0_2} = z_{S_I} + (R_z/R_x) (x_{0_2} - x_{S_I}),$$

$$\text{where } a = 1 + (R_y/R_x)^2$$

$$b = 2y_{S_I} (R_y/R_x) - 2x_{S_I} (R_y/R_x)^2$$

The equations are modified slightly if $R_x = 0$.

The point which lies in the positive direction of the reflection vector is designated (x_o, y_o, z_o) , and the polar coordinates of this point are

$$\alpha = \tan^{-1} (z_o / \sqrt{x_o^2 + y_o^2})$$

$$\beta = \tan^{-1} (y_o / x_o) ,$$

where α is the elevation angle and β is the azimuth angle as shown in Figure 5.

INPUT VARIABLES

The input variables to the computer program are presented in this section.

- TITLE - Title printed at top of each output page.
- DEFLT - Specifies whether default option is to be used during inputting of reflective surface data. The default option reduces the amount of necessary input.

DEFLT = 0 Default option not used

DEFLT = 1 Default option used

- PRINT - Controls how much and what type of data will be printed. Any plots, whether printer plots or CALCOMP plots, will be controlled by another input. Figure 16 shows the results of each option.
- PLOT - Plot options. When plots are requested, a sun glint signature plot for each sun elevation is produced either on CALCOMP plotter or on printer, or both.

PLOT = 0 No plots

PLOT = 1 CALCOMP plots only

PLOT = 2 Print plots only

PLOT = 3 Both CALCOMP and print plots

- PROBL - Specifies whether to calculate probability.

PROBL = 0 Probability calculation bypassed

PROBL = 1 Calculate probability

- THETA - Aircraft pitch attitude in degrees. Positive for nose of aircraft up. Used in forming the inertial-to-body axis transformation shown in Figure 4.

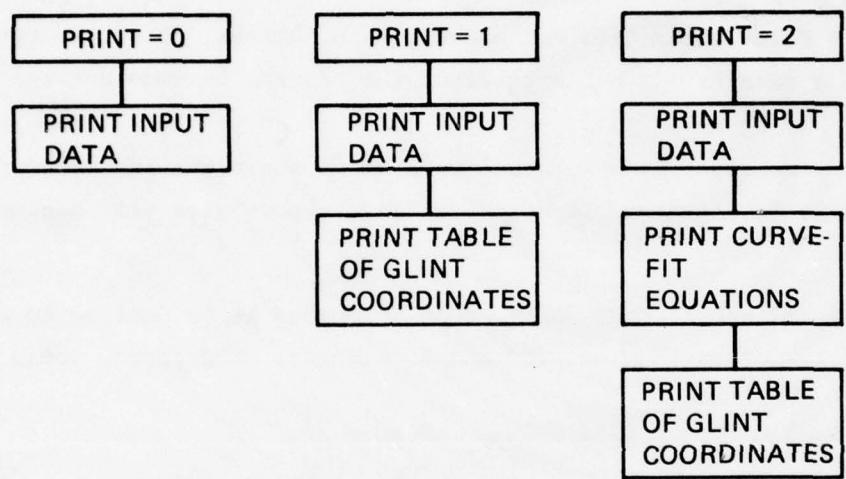


Figure 16. Computer Output From Option PRINT.

- PHI - Aircraft roll attitude in degrees. Positive for right wing down. Used in forming the inertial-to-body axis transformation shown in Figure 4.
- GAMN - Number of different sun elevations to be analyzed. Value should be less than or equal to 9. If not specified, the program assumes a value of 3.
- GAMI - Value of initial sun elevation in degrees. Positive for sun above aircraft. If not specified, the program assumes a value of zero.
- DGAM - Increment by which each successive sun elevation is increased. Value is in degrees. If not specified, the program will assume a value of 30.

Example of use of GAMN, GAMI, DGAM: Suppose it is desired to analyze three sun elevations, 0, 15, and 30 degrees. The inputs would be

GAMN = 3. GAMI = 0. DGAM = 15.

- PSIN - Number of different aircraft headings to be analyzed. Value should be less than or equal to 9. If not specified, a value of 9 is assumed by the program.
- PSII - Value of initial aircraft heading (yaw) in degrees. Heading is positive for nose right as shown in Figure 1. A value of -180 degrees is assumed by the program when no value is specified.
- DPSI - Increment by which each successive heading is increased. Value is in degrees. If no value is specified, a value of 45 degrees is used.
- XROT - Station that locates the center of rotation of the aircraft (see Figure 4). Usually the aircraft center of gravity. If not specified, a value of zero is used.
- YROT - Buttline location of the center of rotation. If not specified, a value of zero is used.

- ZROT - Waterline location of the center of rotation. If not specified, a value of zero is used.
- DISTG - Radial distance (feet) in inertial x-y plane from origin of inertial axes to projected glint. This distance is specified as R in Figure 5. If not specified, a value of 1,000 feet is used by the program.
- ELVMIN - Lower limit on elevation angle of possible observers. Value is in degrees. If not specified, a value of -10 degrees is used. ELVMIN is used in probability calculation and lower boundary on sun glint signature plots. Value cannot be less than -90 degrees.
- ELVMAX - Upper limit on elevation angle of possible observers. Value is in degrees. If not specified, a value of 10 degrees is used. ELVMAX is used in probability calculation and upper boundary on sun glint signature plot. Value cannot be greater than 90 degrees.
- AZMMIN - Minimum allowable azimuth angle to observer, in degrees. If not specified, a value of -180 degrees is used. AZMMIN is the boundary on the left of the sun glint signature plot. Value cannot be less than -180 degrees.
- AZMMAX - Maximum allowable azimuth angle to observer, in degrees. If not specified, a value of 180 degrees is used. AZMMAX is the boundary on the right of the sun glint signature plot. Value cannot be greater than 180 degrees.
- FENCES - Number of fences. Maximum number is 10.
- PANELS - Number of reflective (transparency) panels. Maximum number is 12.
- FPTS - Number of boundary points on a fence. Maximum value is 10.
- XFPT - Station for a fence boundary point.
- YFPT - Buttline for a fence boundary point.

- ZFPT - Waterline for a fence boundary point.
- TITLES - Transparency (reflective) surface identification.
- SYMTRY - Specifies whether the transparency surface just identified is symmetric to the surface, whose data has just been read in. This option will be made clearer with the sample cases in Appendix A. The possible values for SYMTRY are:

SYMTRY = 0. Not symmetric to previous surface and therefore data for this surface is necessary.

SYMTRY = 1. Symmetry is about x-z plane.

SYMTRY = 2. Symmetry is about x-y plane.

SYMTRY = 3. Symmetry is about y-z plane.

If SYMTRY \neq 0, then the following variables are not necessary for this reflective surface.

The variables XFIT, YFIT, ZFIT, XYFIT, XZFIT, YZFIT, XSQFIT, YSQFIT, ZSQFIT, XCOF, YCOF, ZCOF, XYCOF, XZCOF, YZCOF, XSQC0F, YSQCOF, ZSQCOF, and CTERM are used only if DEFLT = 0. With these variables, the program user may either specify which terms in the general quadratic equation he wishes to use in the curve-fit routine, or he may specify the actual values for these coefficients.

- XFIT - Should the linear x term in the quadratic equation be used in the curve-fit analysis of this surface?

XFIT = 0. No

XFIT = 1. Yes

- YFIT - Should the linear y term in the quadratic equation be used in the curve-fit analysis of this surface?

YFIT = 0. No

YFIT = 1. Yes

- ZFIT - Should the linear z term in the quadratic equation be used in the curve-fit analysis of this surface?

ZFIT = 0. No

ZFIT = 1. Yes

- XYFIT - Should the quadratic xy term in the quadratic equation be used in the curve-fit analysis of this surface?

XYFIT = 0. No

XYFIT = 1. Yes

- XZFIT - Should the quadratic xz term in the quadratic equation be used in the curve-fit analysis of this surface?

XZFIT = 0. No

XZFIT = 1. Yes

- YZFIT - Should the quadratic yz term in the quadratic equation be used in the curve-fit analysis of this surface?

YZFIT = 0. No

YZFIT = 1. Yes

- XSQFIT - Should the quadratic x^2 term in the quadratic equation be used in the curve-fit analysis of this surface?

XSQFIT = 0. No

XSQFIT = 1. Yes

- YSQFIT - Should the quadratic y^2 term in the quadratic equation be used in the curve-fit analysis?

YSQFIT = 0. No

YSQFIT = 1. Yes

- ZSQFIT - Should the quadratic z^2 term in the quadratic equation be used in the curve-fit analysis?

ZSQFIT = 0. No

ZSQFIT = 1. Yes

- XCOF - If a nonzero value is read in for XCOF, then the value for XCOF will be used for the coefficient of x in the quadratic equation representing this reflective surface. XFIT must be set to one when XCOF is used.
- YCOF - If a nonzero value is read in for YCOF, then the value for YCOF will be used for the coefficient of y in the quadratic equation representing this reflective surface. YFIT must be set to one when YCOF is used.
- ZCOF - If a nonzero value is read in for ZCOF, then the value for ZCOF will be used for the coefficient of z in the quadratic equation representing this reflective surface. ZFIT must be set to one when ZCOF is used.
- XYCOF - If a nonzero value is read in for XYCOF, then the value for XYCOF will be used for the coefficient of xy in the quadratic equation representing this reflective surface. XYFIT must be set to one when XYCOF is used.
- XZCOF - If a nonzero value is read in for XZCOF, then the value for XZCOF will be used for the coefficient of xz in the quadratic equation representing this reflective surface. XZFIT must be set to one when XZCOF is used.

- XSQCOF - If a nonzero value is read in for XSQCOF, the value for XSQCOF will be used for the coefficient of x^2 in the quadratic equation representing this reflective surface. XSQFIT must be set to one when XSQCOF is used.
- YSQCOF - If a nonzero value is read in for YSQCOF, the value for YSQCOF will be used for the coefficient of y^2 in the quadratic equation representing this reflective surface. YSQFIT must be set to one when YSQCOF is used.
- ZSQCOF - If a nonzero value is read in for ZSQCOF, the value for ZSQCOF will be used for the coefficient of z^2 in the quadratic equation representing the reflective surface. ZSQFIT must be set to one when ZSQCOF is used.
- CTERM - If a nonzero value is read in for CTERM, the value for CTERM will be used for the constant term in the quadratic equation for this reflective surface.
- RPTS - Number of boundary points on reflective surface. Maximum value is 30.
- XSPT - Station for a reflective surface boundary point.
- YSPT - Buttline for a reflective surface boundary point.
- ZSPT - Waterline for a reflective surface boundary point.

Fifty-one input parameters have been defined, but the actual number of inputs depends on the particular case. The arrangement of the input parameters on data cards for a configuration consisting of two fences and two reflective surfaces is shown in Figure 17. The first fence has three boundary points and the second fence has four. The two reflective surfaces are symmetric about the x-z plane and are defined by four boundary

COLUMN 1							COLUMN 80	
TITLE								
1	11	21	31	41	51	60	80	80
DEFLT*	PRINT	THETA	PHI					
1	11	21	31	41	51	60	80	80
GAMN	GAMI	DGAM	PSIN	PSII	DPSI			
1	11	21	31	41	51	60	80	80
XROT	YROT	ZROT						
1	11	21	31	41	51	60	80	80
DISTG	ELVMIN	ELVMAX	AZMMIN	AZMMAX				
1	11	21	31	41	51	60	80	80
FENCES	PANELS	PLOT	PROBL					
1	11	21	31	41	51	60	80	80
FPTS								
1	11	21	31	41	51	60	80	80
XFPT(1)	YFPT(1)	ZFPT(1)	XFPT(2)	YFPT(2)	ZFPT(2)			
1	11	21	31	41	51	60	80	80
XFPT(3)	YFPT(3)	ZFPT(3)						
1	11	21	31	41	51	60	80	80
FPTS								
1	11	21	31	41	51	60	80	80
XFPT(1)	YFPT(1)	ZFPT(1)	XFPT(2)	YFPT(2)	ZFPT(2)			
1	11	21	31	41	51	60	80	80
XFPT(3)	YFPT(3)	ZFPT(3)	XFPT(4)	YFPT(4)	ZFPT(4)			
1	11	21	31	41	51	60	80	80
TITLES								
1	11	21	31	41	51	60	80	80
SYMTRY*								
1	11	21	31	41	51	60	80	80
XFIT	YFIT	ZFIT	XYFIT	XZFIT	YZFIT			
1	11	21	31	41	51	60	80	80
XSQFIT	YSQFIT	ZSQFIT						
1	11	21	31	41	51	60	80	80
XCOF	YCOF	ZCOF	XYCOF	XZCOF	YZCOF			
1	11	21	31	41	51	60	80	80
XSOCOF	YSOCOF	ZSOCOF	CTERM					
1	11	21	31	41	51	60	80	80
RPTS								
1	11	21	31	41	51	60	80	80
XSPT(1)	YSPT(1)	ZSPT(1)	XSPT(2)	YSPT(2)	ZSPT(2)			
1	11	21	31	41	51	60	80	80
XSPT(3)	YSPT(3)	ZSPT(3)	XSPT(4)	YSPT(4)	ZSPT(4)			
1	11	21	31	41	51	60	80	80
XSPT(5)	YSPT(5)	ZSPT(5)						
1	11	21	31	41	51	60	80	80
TITLES								
1	11	21	31	41	51	60	80	80
SYMTRY*								

* FOR THIS SETUP
 DEFLT = 0.
 SYMTRY = 1. (SECOND REFLECTIVE SURFACE)
 ** DEFLT = 1.

Figure 17. Input Data Scheme.

points. The default option will not be used in reading in the reflective surface data; however, the comments show which cards are omitted if the default option is used.

GENERATED SYMBOLS

The symbolic names for the variables of interest which are assigned values during the execution of the program are presented in this section.

- COEFFB - Array containing the coefficients of the equations representing the fences.
- COEFSB - Array containing the coefficients of the equations representing the reflective surfaces.
- SIGF - Array containing the root-mean-square estimate of the error for each fence curve-fit.
- SIGS - Array containing the root-mean-square estimate of the error for each reflective surface curve-fit.
- GAMMA - Sun elevation angle. Takes on values specified by the inputs GAMN, GAMI, and DGAM.
- ALPHA2 - Array containing the elevation angles for all the sun glints generated.
- BETA2 - Array containing the azimuth angles for all the sun glints generated.
- INTRFL - Array containing a set of condition flags. Each flag records whether a particular set of sun glint coordinates was generated by an internal reflection.

INTRFL = 0 External reflection

INTRFL = 1 Internal reflection

- ISHADW - Array containing a set of condition flags. Each flag records whether a particular set of sun glint coordinates was generated from a boundary point in the shadow of a fence.

ISHADW = 0 Boundary point not in shadow

ISHADW = 1 Boundary point in shadow

- INTERF - Array containing a set of condition flags. Each flag records whether a particular set of sun glint coordinates was generated by a reflection vector which intersected a fence.

INTERF = 0 No intersection with fence

INTERF = 1 Reflection vector blocked by a fence

- APLOT - Array containing the minimum and maximum elevation angles which were formed at each aircraft heading.

- BPLOT - Array containing the minimum and maximum azimuth angles which were formed at each aircraft heading.

- SUN - Three-dimensional array containing the coordinates of the unit sun vector in inertial axes.

- SUNB - Three-dimensional array containing the coordinates of the unit sun vector in aircraft body axes.

- DIRCOS - Three-by-three matrix containing the orthonormal transformation shown in Figure 4 used to transform coordinates of a vector from inertial to body axes.

- XB - Coordinates of a reflective surface boundary point in body axes. Array is redefined for each boundary point being analyzed.

- X - Coordinates of a reflective surface boundary point in inertial axes.

- VNORMB - Coordinates of normal to reflective surface.

- RFLTN - Coordinates of reflection vector referred to inertial axes.
- XPLOT - Array of azimuthal values of points forming the polygon representing the area swept by glints from a reflective surface.
- YPLOT - Array of elevation angles for points forming the polygon representing the area swept by glints from a reflective surface.
- SWEET - Area swept by the sun glints from all the reflective surfaces, ignoring overlap.
- SWA2 - Area swept by the sun glints from all the reflective surfaces, counting the overlap area only once.
- TAREA - Total possible area of sun glint signature plot.
- VPROBL - Value of probability calculation. SWA2 divided by TAREA.

OUTPUT

The output from the program is broken up into five sections and controlled by the input variables PRINT and PLOT as defined previously. The five types of output are:

- Input data
- Curve-fit results
- Tabular sun glint results
- Sun glint signature plots using printer
- Sun glint signature plots using CALCOMP plotter.

The printout of the input data is just a feedout of the data read in from cards. Sheet 1 of Figure 18 shows the first page of this type output. The first group of input data is printed out under the caption **CONTROL OPTIONS. The input parameters presented are DEFLT, PRINT, PLOT, and PROBL, respectively. The second group of input data comes under the caption **AIRCRAFT INITIAL EULER ORIENTATION. The input parameters printed are THETA, PHI, AND PSII. Under the third and fourth captions, **SELECTED YAW ANGLE ROTATIONS and **SELECTED SUN ELEVATIONS, are printed the values for the variables PSIN, PSII, DPSI, and GAMN, GAMI, DGAM, respectively. The fifth set of information, **REFERENCE INFORMATION FOR MEASURING SUN GLINT ANGLES, in the printout of XROT, YROT, ZROT, and DISTG. The sixth and seventh sets of information, **PLOT SCALING, and **BOUNDARIES FOR CALCULATING PROBABILITY, are printouts of the same variables, AZMMIN, AZMMAX, ELVMIN, and ELVMAX. They are printed out twice, under different titles, to show that the same values are used in both places. Sheet 2 of Figure 18 is a printout of the fence input data. The number of fences (FENCES), the number of boundary points (FPTS), and the location of each boundary point (XFPT, YFPT, ZFPT) are printed out. Sheet 3 of Figure 18 shows how the input data for a reflection surface is output. Under **REFLECTIVE SURFACES DATA, the number of reflective panels (PANELS) is printed. Then, for each reflective surface, an identification title (TITLES), preselected curve-fit terms (XFIT, YFIT, ZFIT, XYFIT, XZFIT, YZFIT, XSQFIT, YSQFIT, ZSQFIT),

```

*** INPUT DATA ***

*CONTROL OPTIONS
  DEFAULT OPTION 0.0 PRINT OPTION 2.
  PLOT OPTION 1. PROBABILITY OPT 1.

*AIRCRAFT INITIAL EULER ORIENTATION
  PITCH -3.8 ROLL 0.0 YAW 0.0

**SELECTED TAN ANGLE ROTATIONS
  NO. OF ANGLES 2. INITIAL ANGLE 0.0 ANGLE INCREMENT 20.0

**SELECTED SUN ELEVATIONS
  NO. OF ANGLES 2. INITIAL ANGLE 0.0 ANGLE INCREMENT 30.0

*REFERENCE INFORMATION FOR MEASURING SUN GLINT ANGLES
  REFERENCE POINT FOR MEASURING ANGLES OF GLINT SIGNATURE DISTANCE AT WHICH GLINT SIGNATURE FORMED (FT.) 66,000

  X 0.0 Y 0.0 Z 0.0

*PLOT SCALING
  GLINT AZIMUTH GLINT ELEVATION
  MINIMUM -180.0 MINIMUM -10.0
  MAXIMUM 180.0 MAXIMUM 10.0

*BOUNDARIES FOR CALCULATING PROBABILITY
  GLINT AZIMUTH GLINT ELEVATION
  MINIMUM -180.0 MINIMUM -10.0
  MAXIMUM 180.0 MAXIMUM 10.0

```

Figure 18. Typical Printout of Input Data (Sheet 1 of 3).

*** INPUT DATA ***					
FENCE DATA					No. OF FENCES
1.					
*FENCE NO. 1					
NUMBER OF POINTS 4.					
BOUNDARY POINTS -- BODY AXES (STATION LINE, BUTTLINE, WATERLINE)					
	X	Y	Z	X	Y
(1)	-132.00	0.0	-12.00	(2)	-132.00
(4)	-48.00	0.0	-12.00		
				22.56	
					0.0
				(3)	-48.00
					22.56

Figure 18. Typical Printout of Input Data (Sheet 2 of 3).

```

** INPUT DATA **

*REFLECTIVE SURFACE DATA
NO. OF SURFACES 4.

*REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD.

CURVE-FIT INPUT DATA
DESIRED TERMS FOR CURVE-FITTING
(1) X TERM 1. (2) Y TERM 1. (3) Z TERM 1. (4) XY TERM 0. (5) XZ TERM 0.
(6) YZ TERM 0. (7) YSQ TERM 0. (8) YSSQ TERM 0. (9) ZSQ TERM 0.

INPUT VALUES FOR COEFFICIENTS OF SURFACE
(11) X COEF 0.0 (12) Y COEF 0.0 (13) Z COEF 0.0
(41) XY COEF 0.0 (51) XZ COEF 0.0 (61) YZ COEF 0.0
(71) YSQ COEF 0.0 (81) YSSQ COEF 0.0 (91) ZSQ COEF 0.0
(101) CONSTANT 0.0

POUNDRY POINTS -- BODY AXES (STATIONLINE+WATERLINE)
X Y
(1) -15.44 2 -18.00 1 2 -153.48 10.00 2 -12.00 1 3 -132.00 10.00 2
(4) -15.44 -18.00 16.56 16.56 16.56 16.56 16.56 16.56 16.56

*REFLECTIVE SURFACE NO. 2 IDENTIFICATION: TOP PANEL

CURVE-FIT INPUT DATA
DESIRED TERMS FOR CURVE-FITTING
(1) X TERM 1. (2) Y TERM 1. (3) Z TERM 1. (4) XY TERM 0. (5) XZ TERM 0.
(6) YZ TERM 0. (7) YSQ TERM 0. (8) YSSQ TERM 0. (9) ZSQ TERM 0.

INPUT VALUES FOR COEFFICIENTS OF SURFACE
(11) X COEF 0.0 (12) Y COEF 0.0 (13) Z COEF 0.0
(41) XY COEF 0.0 (51) XZ COEF 0.0 (61) YZ COEF 0.0
(71) YSQ COEF 0.0 (81) YSSQ COEF 0.0 (91) ZSQ COEF 0.0
(101) CONSTANT 0.0

POUNDRY POINTS -- BODY AXES (STATIONLINE+WATERLINE)
X Y
(1) -152.00 2 -18.00 1 2 -132.00 10.00 2 -16.56 1 3 -48.00 10.00 2
(4) -48.00 -18.00 16.56 16.56 16.56 16.56 16.56 16.56 16.56

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: LEFT FORWARD SIDE PANEL

CURVE-FIT INPUT DATA
DESIRED TERMS FOR CURVE-FITTING
(1) X TERM 1. (2) Y TERM 1. (3) Z TERM 1. (4) XY TERM 0. (5) XZ TERM 0.
(6) YZ TERM 0. (7) YSQ TERM 0. (8) YSSQ TERM 0. (9) ZSQ TERM 0.

```

Figure 18. Typical Printout of Input Data (Sheet 3 of 3).

preselected curve-fit coefficients (XCOF, YCOF, ZCOF, XYCOF, XZCOF, YZCOF, XSQCOF, YSQCOF, ZSQCOF, CTERM), and boundary points are printed.

The second type of output is the result of curve-fitting the boundary points of each fence and reflective surface. Sheet 1 of Figure 19 shows the coefficients of the plane best approximating the boundary points of the fence. For the case shown, the equation would be $0.01476y = 0$ or just $y = 0$, which is the equation of the x-z plane. Sheet 2 of Figure 19 shows the coefficients of several quadratic equations approximating several reflective surfaces. For the case shown, the equation for the first surface is a plane, $-0.006923x + 0.005206z - 1 = 0$.

The third type of output is the tabular sun glint results. Each table represents the results for one reflective surface at constant sun elevation as aircraft heading changes. Part of a typical table is shown in Figure 20. At the top of the page, the reflective surface is identified. Next, the sun elevation, aircraft pitch and roll attitude, location of aircraft center of rotation, and radial distance to observers are printed. Then, for each aircraft heading, each boundary point is listed along with three condition flags and the point on the sun glint resulting from the reflection from this boundary point. The condition flags INRFL, ISHAD, and INTRF specify whether the reflection from this boundary point is internal, whether the boundary point lies in a shadow, or whether the reflection vector from this boundary point intersects a fence. A value of zero means no while a one means yes.

The fourth type of output is a print plot of a sun glint signature, as shown in Figure 21. Each plot is for a constant sun elevation. The values that are plotted are the minimum and maximum azimuths and elevations for each sun glint resulting from each reflective surface at each aircraft heading analyzed. Each letter, as the legend below the plot shows, is for a constant aircraft heading and may have resulted from more than one surface. The probability as previously defined is also shown in the right-hand corner below the plot.

*** RESULTS OF CURVE-FIT ANALYSIS ***

• FENCE DATA

• EERCE NO. 1

VALUES FOR COEFFICIENTS OF SURFACE

111 X COEF 0.1
112 Y COEF 0.15954281
113 Z COEF 0.0
CONSTANT 0.0

ROOT MEAN SQUARE ERROR OF FIT: 0.0

Figure 19. Typical Printout of Curve-Fit Results (Sheet 1 of 2).

*** RESULTS OF CURVE-FIT ANALYSIS ***

*REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD

VALUES FOR COEFFICIENTS OF SURFACE

(11) X COFF -0.6223E-02	(21) Y COFF 0.0	(31) Z COFF 0.5226E-12
(41) XY COFF 0.0	(51) XZ COFF 0.0	(61) YZ COFF 0.0
(71) YZ COFF 0.0	(81) YSO COFF 0.0	(91) ZSO COFF 0.0
(101) CONSTANT -0.1980E 01		

ROOT MEAN SQUARE ERROR OF FIT: 0.1688E-15

*REFLECTIVE SURFACE NO. 2 IDENTIFICATION: TOP PANEL

VALUES FOR COEFFICIENTS OF SURFACE

(11) X COFF -0.2020E-17	(21) Y COFF 0.0	(31) Z COFF 0.6039E-01
(41) XY COFF 0.0	(51) XZ COFF 0.0	(61) YZ COFF 0.0
(71) YZ COFF 0.0	(81) YSO COFF 0.0	(91) ZSO COFF 0.0
(101) CONSTANT -0.1000E 01		

ROOT MEAN SQUARE ERROR OF FIT: 0.1171E-15

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: LEFT FORWARD SIDE PANEL

VALUES FOR COEFFICIENTS OF SURFACE

(11) X COFF -0.2007E-15	(21) Y COFF -0.5556E-01	(31) Z COFF 0.7201E-16
(41) XY COFF 0.0	(51) XZ COFF 0.0	(61) YZ COFF 0.0
(71) YZ COFF 0.0	(81) YSO COFF 0.0	(91) ZSO COFF 0.0
(101) CONSTANT -0.1000E 01		

ROOT MEAN SQUARE ERROR OF FIT: 0.1721E-14

*REFLECTIVE SURFACE NO. 4 IDENTIFICATION: LEFT LATERAL SIDE PANEL

VALUES FOR COEFFICIENTS OF SURFACE

(11) X COFF 0.3296E-17	(21) Y COFF -0.5556E-01	(31) Z COFF -0.5363E-18
(41) XY COFF 0.0	(51) XZ COFF 0.0	(61) YZ COFF 0.0
(71) YZ COFF 0.0	(81) YSO COFF 0.0	(91) ZSO COFF 0.0
(101) CONSTANT -0.1000E 01		

ROOT MEAN SQUARE ERROR OF FIT: 0.1695E-15

Figure 19. Typical Printout of Curve-Fit Results (Sheet 2 of 2).

444 SUN GLINT SIGNATURE 000									
REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD			REFLECTIVE SURFACE NO. 2 IDENTIFICATION: WINDSHIELD						
SUN ELEVATION	61.00	A/C PITCH ATTITUDE	-9.00	A/C YAW	0.0	STATION LINE COORDINATES	BODY AXES	STATUS FLAG FOR POINTS	GLINT ELEVATION
A/C PITCH ATTITUDE	0.0	A/C YAW	0.0	STATION LINE COORDINATES	BODY AXES	STATUS FLAG FOR POINTS	GLINT ELEVATION	AZIMUTH	
0.0	-153.48	16.00	-12.00	0	0	0	0	3.05	-1.34
0.0	-153.48	16.00	-12.00	0	0	0	0	3.05	1.34
0.0	-132.00	16.00	16.56	0	0	0	0	5.42	1.34
0.0	-132.00	16.00	16.56	0	0	0	0	5.42	-1.34
20.00	-153.48	-16.00	-12.00	0	0	0	0	1.79	26.60
20.00	-153.48	16.00	-12.00	0	0	0	0	1.77	29.25
20.00	-132.00	16.00	16.56	0	0	0	0	4.18	29.50
20.00	-132.00	-16.00	16.56	0	0	0	0	4.13	26.85
40.00	-153.48	-16.00	-12.00	0	0	0	0	-1.88	53.82
40.00	-153.48	16.00	-12.00	0	0	0	0	-1.88	56.37
40.00	-132.00	16.00	16.56	0	0	0	0	0.35	56.83
40.00	-132.00	-16.00	16.56	0	0	0	0	0.35	54.29
60.00	-153.48	-16.00	-12.00	0	0	0	0	-7.63	79.70
60.00	-153.48	16.00	-12.00	0	0	0	0	-7.51	62.13
60.00	-132.00	16.00	16.56	0	0	0	0	-5.97	82.76
60.00	-132.00	-16.00	16.56	0	0	0	0	-5.68	88.35
80.00	-153.48	-16.00	-12.00	0	0	0	0	-10.98	103.81
80.00	-153.48	16.00	-12.00	0	0	0	0	-10.68	106.13
80.00	-132.00	16.00	16.56	0	0	0	0	-13.92	106.88
90.00	-132.00	-16.00	16.56	0	0	0	0	-13.24	104.56

Figure 20. Typical Printout of Tabular Sun Glint Results.

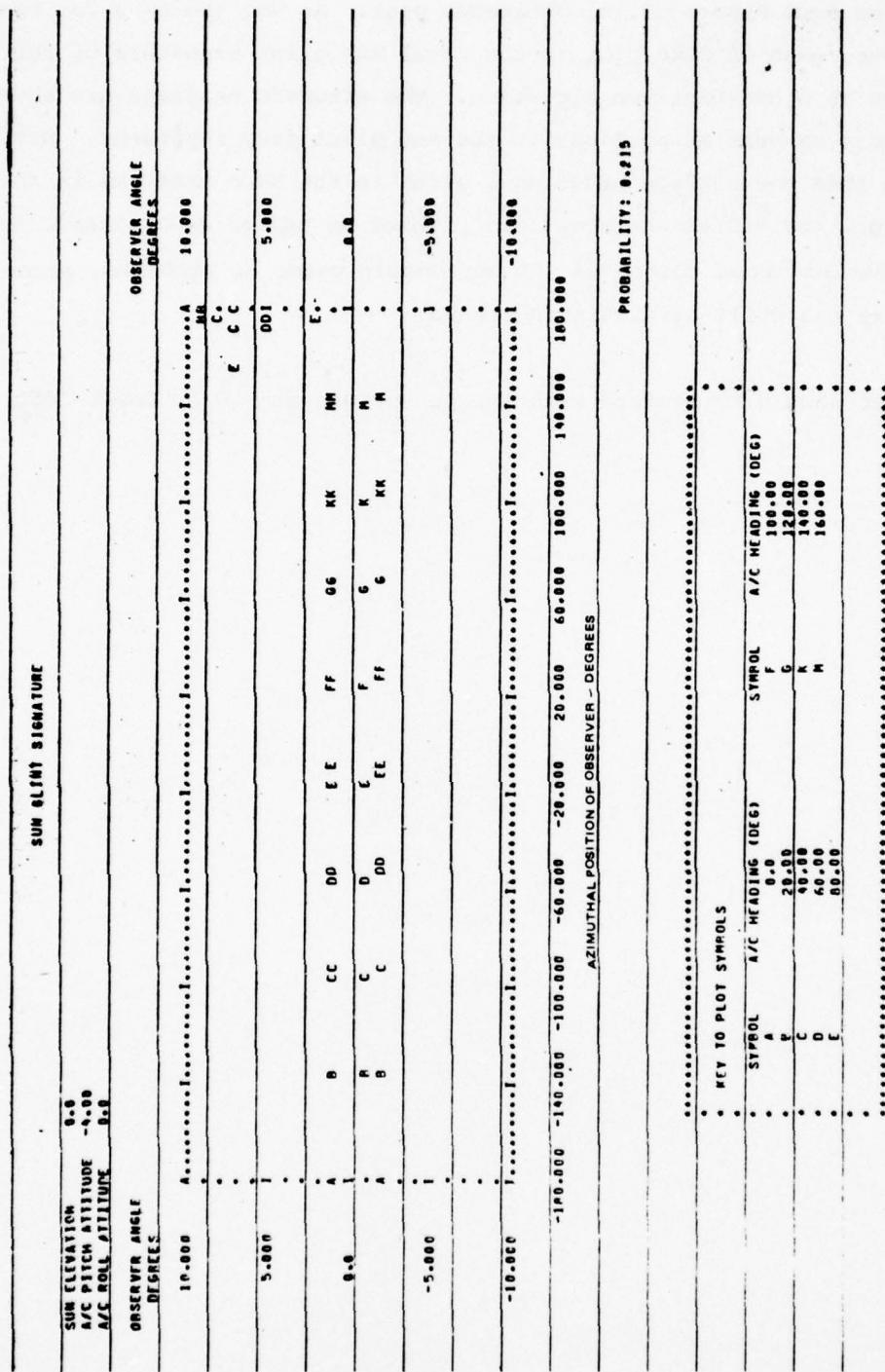


Figure 21. Typical Printout of On-Line Plot.

The fifth and final type of output comes from a CALCOMP plotter. It is, as can be seen from Figure 22, a continuous plot. As was the case for the print plots, each CALCOMP plot is the total sun glint signature of the configuration at a constant sun elevation. The aircraft headings are shown on the plots as near as possible to the sun glint they represent. However, when more than one surface produces a glint in the same area, as in the bottom plot, the values are sometimes printed on top of each other. This nuisance has not been corrected yet but should cause no problems, since the numbers can still be distinguished.

The plotter should be started with the pen origin at 0,0 or Lower Left.

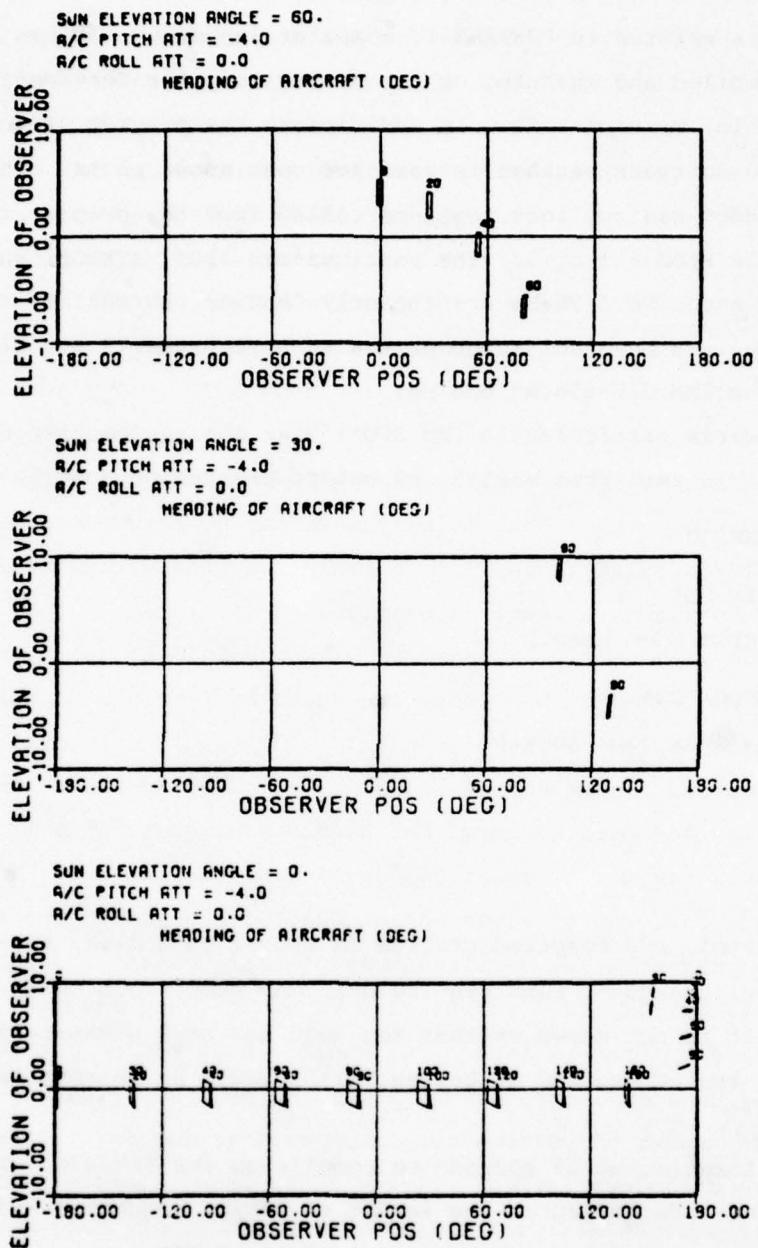


Figure 22. Typical CALCOMP Plot.

RUNNING INSTRUCTIONS

The program is written in FORTRAN IV computer language. It has been successfully compiled and executed on the Boeing Computer Service's IBM 370-158 at Philadelphia, Pennsylvania. In addition to the program itself, the CALCOMP Basic Software package is used for continuous plots. This CALCOMP package includes six routines that are called from the program if the input variable PLOT = 1 or 3. The routines are PLOT, SYMBOL, NUMBER, SCALE, AXIS, and LINE. These are the only CALCOMP subroutines required. The following is a physical setup of the deck required to compile and execute on the IBM 370-158 at Boeing.

```
// JCL cards particular to IBM 370-158 at Boeing Computer Services.  
Input is read from unit 5 and output printed on unit 6.  
  
//EXEC COMP2GO  
  
//FTC.SYSIN DD*  
        (Program goes here.)  
  
//G0.FT05F001 DD*  
        (Data deck goes here.)  
  
/*  
//
```

At Boeing Vertol, the compiled program is stored on a disk; therefore, once compiled, any subsequent runs require only the proper JCL cards and the data deck. It is not known whether the user has such a capability, so any reference to the JCL's used at Boeing Vertol could be meaningless.

The program takes about 35 seconds to compile on the IBM 370. Typical execution times, depending on the amount of output requested and the number of conditions analyzed, are approximately 8 to 12 seconds.

As for the form of the input, all titles are read using an alphanumeric format, which allows for the mixing of letters and numbers in the input. The rest of the inputs are read using a floating-point format with 10

columns allotted for each variable. As long as the decimal point is included, the variable may appear anywhere in those 10 columns. The specific location of each variable on the card was shown in Figure 17.

APPENDIX A
SAMPLE CASES

Two sample cases will be discussed to demonstrate the use of the program. The first example deals with the configuration shown in Figure A-1. The panels are all flat. There are six panels, four of them being identified in the isometric. The right-forward and the right-aft panels are hidden from view. The right-forward panel is symmetric to the left-forward panel about the x-z plane of the aircraft. Likewise the right-aft panel is symmetric to the left-aft panel about the x-z plane of the aircraft. There are two fences on the canopy, one over the windshield while the other is over the top panel. Since the surfaces are all flat, as few as three points can be used to locate the boundaries of the surface. The circled letters represent the boundary points selected. In practice, more boundary points may be selected; however, in most cases of flat panels a large number of boundary points is not necessary. Using the three-view drawing shown, the center of rotation and each circled letter can be located referenced to the station, buttline, waterline axes. The panels and fences can then be described by a set of locations as tabulated in Table A-1. For instance, the windshield is defined by the locations of (A) (B) (C) (D). These are the boundary points that are used by the program.

With the panels and fences defined and the location of the center of rotation known, the next step is to decide on what conditions to analyze. Using Figure 17 as a guide, the first input card would contain the title for this run. The title chosen is FLAT PANELS WITH FENCES SAMPLE CASE I, which will appear on the first card. The second card contains the input variables, DEFLT, PRINT, THETA, and PHI. The default option is chosen (DEFLT = 1.); therefore the optional cards shown in Figure 17 will be omitted. Since all the printout possible is desired, PRINT will be set equal to 2. The aircraft pitch and roll attitudes will be assumed to be zero (THETA = 0. and PHI = 0.). This takes care of the second card. The third card will specify the desired sun elevations and aircraft headings (GAMN, GAMI, DGAM, PSIN, PSII, DPSI). Three sun elevations are chosen, 0, 30, and 60 degrees. Therefore, GAMN will be equal to 3., GAMI will be

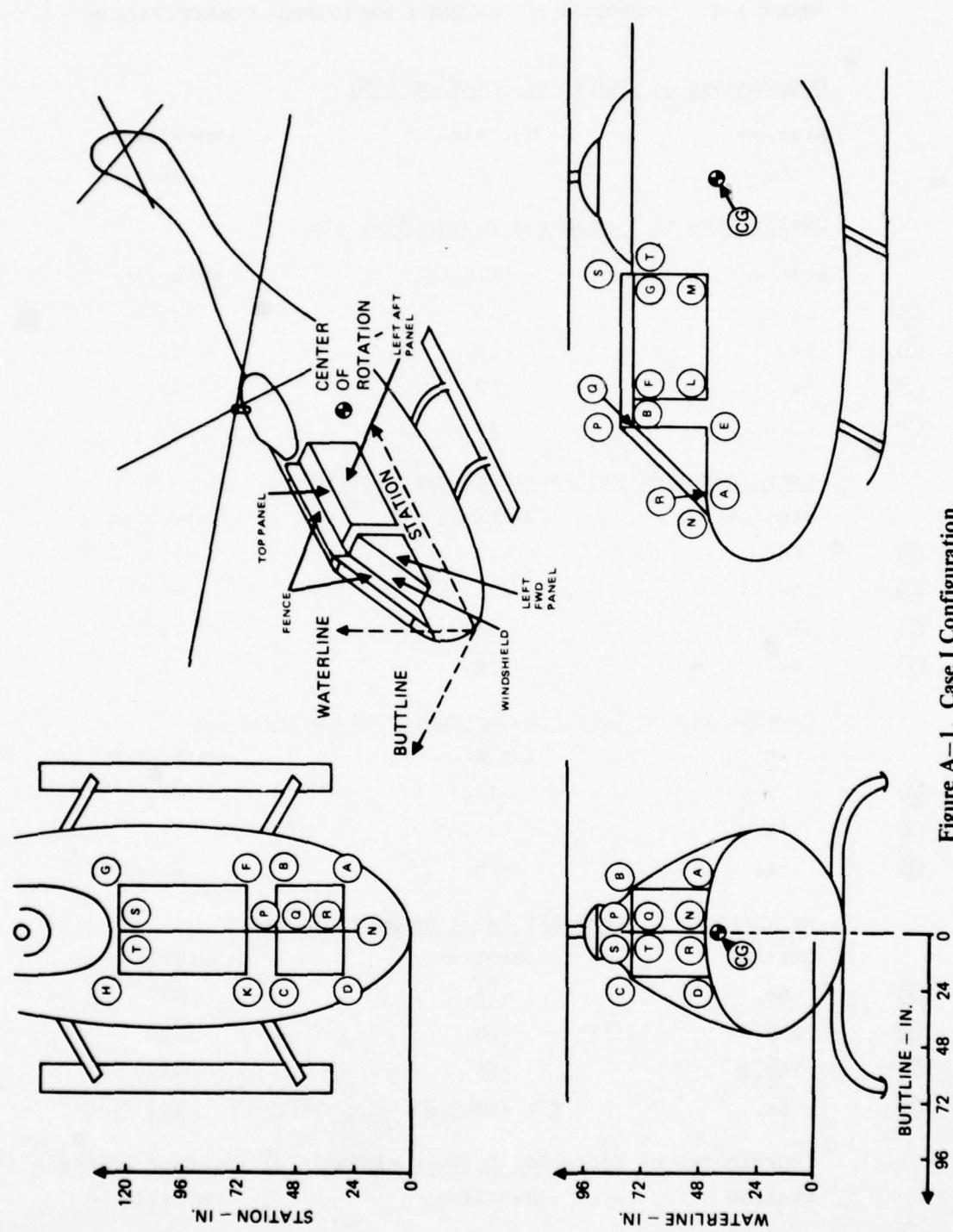


Figure A-1. Case I Configuration.

TABLE A-1. COORDINATES OF CASE I HELICOPTER CONFIGURATION

Coordinates of Center of Rotation (CG)

Station	Buttline	Waterline
150.	0	36.

Coordinates of Windshield Boundary Points

Station	Buttline	Waterline
(A) 27.6	-18.	42.
(B) 54.	-18.	72.
(C) 54.	18.	72.
(D) 27.6	18.	42.

Coordinates of Top Panel Boundary Points

Station	Buttline	Waterline
(F) 66.	-18.	72.
(G) 118.8	-18.	72.
(H) 118.8	18.	72.
(K) 66.	18.	72.

Coordinates of Left Forward-Panel Boundary Points

Station	Buttline	Waterline
(A) 27.6	-18.	42.
(B) 54.	-18.	72.
(E) 54.	-18.	42.

Coordinates of Left-Aft Panel Boundary Points

Station	Buttline	Waterline
(F) 66.	-18.	72.
(G) 118.8	-18.	72.
(M) 118.8	-18.	42.
(L) 66.	-18.	42.

Coordinates of Fence No. 1 (Over Windshield) Boundary Points

Station	Buttline	Waterline
(N) 22.32	0	42.
(P) 54.	0	78.

TABLE A-1 - Continued

Coordinates of Fence No. 1 - Continued

Station	Buttline	Waterline
② 54.	0	72.
③ 27.6	0	42.

Coordinates of Fence No. 2 (Over Top Panel) Boundary Points

Station	Buttline	Waterline
④ 54.	0	78.
⑤ 118.8	0	78.
⑥ 118.8	0	72.
⑦ 54.	0	72.

0., and DGAM will be 30. Aircraft headings shall cover the whole 360-degree range; therefore, PSIN will be equal to 9., PSII = -180., and DPSI = 45. These six values will take care of the third card. The fourth card contains the location of the center of rotation (XROT, YROT, ZROT). These values are measured on Figure A-1 and recorded at the top of Table A-1. The values are XROT = 150., YROT = 0., and ZROT = 36.

This is all the information needed for the fourth card. The fifth card contains the radial distance to all observers (DISTG), the minimum (ELVMIN) and maximum (ELVMAX) elevation angles, and the minimum (AZMMIN) and maximum (AZMMAX) azimuth angles. The distance to the observer will be assumed to be 1,000 feet. The elevation limits on the observers will be ± 10 degrees and the azimuth limits will be ± 180 degrees. Therefore the fifth card has DISTG = 1,000 feet, ELVMIN = -10 degrees, ELVMAX = 10 degrees, AZMMIN = -180 degrees and AZMMAX = 180 degrees. The sixth card contains the number of fences (FENCES), the number of reflective panels (PANELS), the plot option chosen (PLOT), and whether to calculate probability. For this configuration, FENCES = 2. and PANELS = 6. To get both printer-type and CALCOMP-type plots, set PLOT = 3., and to have probability calculated, PROBL = 1. This takes care of the control cards. The next six cards will contain fence data. The seventh card contains the number of boundary points associated with the first fence (over the windshield). There are four points; therefore, on the seventh card FPTS = 4. The eighth and ninth cards contain the four boundary points (N) (P) (Q) (R) from Table A-1.

The tenth card has the value for the number of boundary points defined for the second fence (over the top panel), FPTS = 4. The eleventh and twelfth cards contain the four boundary points (P) (S) (T) (Q). This completes the fence data. The rest of the data cards contain information about the reflective surfaces. The thirteenth card is an identification card for the first reflective surface read in. For this case it is the WINDSHIELD. The fourteenth card is the symmetry card for this surface. Since this is the first surface and therefore data must be read in for it, SYMTRY = 0. Since DEFLT = 1., the optional cards shown in Figure 17 pertaining to selecting

certain terms in the curve-fit equation will be omitted. The fifteenth card contains RPTS, the number of boundary points for this surface; RPTS has a value of 4. The sixteenth and seventeenth cards contain the four boundary points for this surface, \textcircled{A} \textcircled{B} \textcircled{C} \textcircled{D} .

This completes the data for the first reflective surface. The next surface to be read in will be the TOP PANEL, so identified on the eighteenth card. The nineteenth card is the symmetry card for this panel. This panel is not symmetric in any way to the windshield; therefore SYMTRY = 0. for this surface. The twentieth card contains the number of boundary points for this panel (RPTS = 4.). The twenty-first and twenty-second cards are for the four boundary points \textcircled{F} \textcircled{G} \textcircled{H} \textcircled{K} . This completes the data for the second reflective surface. The next surface to be read in is the LEFT FWD PANEL, so identified on the twenty-third card. This panel is symmetric to the right forward panel, but since it is read in first, SYMTRY = 0 for this surface. The twenty-fourth card is for the number of boundary points recorded for the left-forward surface; this number is three. The twenty-fifth and twenty-sixth cards are used for the three boundary points for this surface, \textcircled{A} \textcircled{B} \textcircled{E} . The next panel to be read in should be the RIGHT FWD PANEL, since it is symmetric to the panel just read in, and therefore only an identification card and symmetry card are required. The twenty-seventh card will contain the identification for the right-forward panel, while the twenty-eighth card is the symmetry card which has a value of one since the right-and left-forward surfaces are symmetric about the x-z plane. The next panel to be read in is the LEFT AFT PANEL, so identified on the twenty-ninth card. Since this panel is not symmetric to the right-forward panel just read in, the symmetry card (the thirtieth card) contains a zero. The thirty-first card represents the number of boundary points for the left-aft panel (RPTS = 4.). The thirty-second and thirty-third cards contain the four boundary points for this surface, \textcircled{F} \textcircled{G} \textcircled{M} \textcircled{L} . The next surface to be read in is the RIGHT AFT PANEL, so identified on the thirty-fourth card. Since this surface is symmetric to the surface just read in, namely the left-aft panel, the thirty-fifth card (the symmetry card) will have a value of one, representing symmetry about the x-z

plane. This completes the input data for this case. The complete input in card image form is shown in Figure A-2. The complete results from this input are shown on pages 71 through 116.

FLAT SURFACES WITH FENCES			SAMPLE CASE ONE		
1.	2.	0.	0.		
3.	0.	30.	9.	-180.	45.
150.	0.	36.			
1000.	-10.	10.	-180.	180.	
2.	6.	3.	1.		
4.					
22.32	0.	42.	54.	0.	72.
54.	0.	72.	27.6	0.	42.
4.					
54.	0.	78.	118.8	0.	72.
118.8	0.	72.	54.	0.	72.
WINDSHIELD					
0.					
4.					
27.6	-18.	42.	54.	-18.	72.
54.	18.	72.	27.6	18.	42.
TOP PANEL					
0.					
4.					
56.	-18.	72.	118.8	-18.	72.
118.8	18.	72.	66.	18.	72.
LEFT FORWARD PANEL					
0.					
3.					
27.6	-18.	42.	54.	-18.	72.
54.	-18.	42.			
RIGHT FORWARD PANEL					
1.					
LEFT AFT PANEL					
0.					
4.					
56.	-18.	72.	118.8	-18.	72.
118.8	-18.	42.	66.	-18.	42.
RIGHT AFT PANEL					
1.					

Figure A-2. Input for Case I; Card Images.

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** INPUT DATA ***

** CONTROL OPTIONS
 DEFAULT OPTION 1.
 PLOT OPTION 3.
 PRINT OPTION 2.
 PROBABILITY NOT 1.

** AIRCRAFT INITIAL EULER ORIENTATION
 PITCH 0.0
 ROLL 0.0
 YAW -150.0

** SELECTED YAW ANGLE ROTATIONS
 NO. OF ANGLES 9.
 INITIAL ANGLE -180.0
 ANGLE INCREMENT 45.0

** SELECTED SUN ELEVATIONS
 NO. OF ANGLES 3.
 INITIAL ANGLE 0.0
 ANGLE INCREMENT 30.0

** REFERENCE INFORMATION FOR MEASURING SUN GLINT ANGLES
 REFERENCE POINT FOR MEASURING ANGLES OF GLINT SIGNATURE
 X 150.00
 Y 0.0
 Z 36.00

** PLOT SCALING
 GLINT AZIMUTH -180.0
 MINIMUM -180.0
 MAXIMUM 180.0
 GLINT ELEVATION
 MINIMUM -10.0
 MAXIMUM 10.0

** BOUNDARIES FOR CALCULATING PROBABILITY
 GLINT AZIMUTH CLINT ELEVATION
 MINIMUM -180.0
 MAXIMUM 180.0
 CLINT AZIMUTH
 MINIMUM -15.0
 MAXIMUM 15.0

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

*** INPUT DATA ***

** FENCE DATA
 NO. OF FENCES 2.

* FENCE NO. 1
 NUMBER OF POINTS 4.

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)
 (1) 22.32 0.0 42.00 (2) 50.00 0.0 78.00 (3) 54.00 0.0 72.00
 (4) 27.66 0.0 42.00

* FENCE NO. 2
 NUMBER OF POINTS 4.

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)
 (1) 54.00 0.0 78.00 (2) 116.80 0.0 78.00 (3) 118.60 0.0 72.00
 (4) 54.00 0.0 72.00

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

*** INPUT DATA ***

**REFLECTIVE SURFACE DATA
NO. OF SURFACES 6.

*REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0. (2) Y TERM 0. (3) Z TERM 0. (4) XY TERM 0. (5) XZ TERM 0. (6) YZ TERM 0. (7) XSG TERM 0. (8) YSG TERM 0. (9) ZSG TERM 0.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0 (2) Y COEF 0.0 (3) Z COEF 0.0 (4) XY COEF 0.0 (5) XZ COEF 0.0 (6) YZ COEF 0.0 (7) XSG COEF 0.0 (8) YSG COEF 0.0 (9) ZSG COEF 0.0 (10) CONSTANT 0.0

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

(1) X 27.60 Y 2 Z 2 V 2
(2) X 27.60 Y 42.00 Z 20 V 2
(3) X 18.00 Y 42.00 Z 20 V 2
(4) X 18.00 Y 42.00 Z 20 V 2
(5) X 18.00 Y 42.00 Z 20 V 2
(6) X 18.00 Y 42.00 Z 20 V 2
(7) X 18.00 Y 42.00 Z 20 V 2
(8) X 18.00 Y 42.00 Z 20 V 2
(9) X 18.00 Y 42.00 Z 20 V 2
(10) X 18.00 Y 42.00 Z 20 V 2

*REFLECTIVE SURFACE NO. 2 IDENTIFICATION: TOP PANEL

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0. (2) Y TERM 0. (3) Z TERM 0. (4) XY TERM 0. (5) XZ TERM 0. (6) YZ TERM 0. (7) XSG TERM 0. (8) YSG TERM 0. (9) ZSG TERM 0.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0 (2) Y COEF 0.0 (3) Z COEF 0.0 (4) XY COEF 0.0 (5) XZ COEF 0.0 (6) YZ COEF 0.0 (7) XSG COEF 0.0 (8) YSG COEF 0.0 (9) ZSG COEF 0.0 (10) CONSTANT 0.0

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

(1) X 66.00 Y 2 Z 2 V 2
(2) X 66.00 Y 22.00 Z 20 V 2
(3) X 66.00 Y 22.00 Z 20 V 2
(4) X 66.00 Y 22.00 Z 20 V 2
(5) X 66.00 Y 22.00 Z 20 V 2
(6) X 66.00 Y 22.00 Z 20 V 2
(7) X 66.00 Y 22.00 Z 20 V 2
(8) X 66.00 Y 22.00 Z 20 V 2
(9) X 66.00 Y 22.00 Z 20 V 2
(10) X 66.00 Y 22.00 Z 20 V 2

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: LEFT FORWARD PANEL

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0. (2) Y TERM 0. (3) Z TERM 0. (4) XY TERM 0. (5) XZ TERM 0. (6) YZ TERM 0. (7) XSG TERM 0. (8) YSG TERM 0. (9) ZSG TERM 0.

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: LEFT FORWARD PANFL

*** INPUT DATA ***

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	5.0	(2) Y COEF	0.0	(3) Z COFF	0.0
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSQ COEF	0.0	(8) YSQ COEF	0.0	(9) ZSC COFF	0.0
(10) CONSTANT	0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z
(1)	27.60	-18.00
	42.00	
		(2)
		54.00
		-18.00
		72.00

*REFLECTIVE SURFACE NO. 4 IDENTIFICATION: RIGHT FORWARD PANEL

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM	0.	(2) Y TERM	0.	(3) Z TERM	0.	(4) XY TERM	0.	(5) XZ TERM	0.
(6) YZ TERM	0.	(7) YSQ TERM	0.	(8) YSC TERM	0.	(9) ZSC TERM	0.		

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.0	(2) Y COEF	0.0	(3) Z COFF	0.0
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSQ COEF	0.0	(8) YSQ COEF	0.0	(9) ZSC COFF	0.0
(10) CONSTANT	0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z
(1)	27.60	18.00
	42.00	
		(2)
		54.00
		18.00
		72.00

*REFLECTIVE SURFACE NO. 5 IDENTIFICATION: LEFT AFT PANFL

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TFRM	0.	(2) Y TERM	0.	(3) Z TERM	0.	(4) XY TERM	0.	(5) XZ TERM	0.
(6) YZ TERM	0.	(7) YSQ TERM	0.	(8) YSC TERM	0.	(9) ZSC TERM	0.		

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.0	(2) Y COEF	0.0	(3) Z COFF	0.0
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSQ COEF	0.0	(8) YSQ COEF	0.0	(9) ZSC COFF	0.0
(10) CONSTANT	0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z
(1)	66.00	-18.00
	72.00	
		(2)
		118.80
		-18.00
		72.00

X	Y	Z
(4)	66.00	-18.00
	42.00	

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

•REFLECTIVE SURFACE NO. 6 IDENTIFICATION: RIGHT AFT PANEL

*** INPUT DATA ***
 CURVE-FIT INPUT DATA
 DESIRED TERMS FOR CURVE-FITTING
 (1) X TERM 0. (2) Y TERM 0.
 (3) Z TERM 0. (4) XY TERM 0.
 (5) XZ TERM 0. (6) YZ TERM 0.
 (7) XSG TERM 0. (8) ZSG TERM 0.
 (9) YSG TERM 0.

INPUT VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COFF 0.0 (2) Y COFF 0.0 (3) Z COFF 0.0
 (4) XY COFF 0.0 (5) XZ COFF 0.0 (6) YZ COFF 0.0
 (7) XSG COFF 0.0 (8) YSG COFF 0.0 (9) ZSG COFF 0.0
 (10) CONSTANT 0.0

BOUNDARY POINTS -- BODY AXES (STATION,LINE,BUTTLINE,WATERLINE)
 X Y Z
 (1) 66.00 1A.01 72.00 (2) 11A.80 1B.00 72.00
 (4) 66.00 1A.00 42.00

WARNING TEST IN SIGN SHOWS VECTORS PERPENDICULAR

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

*** RESULTS OF CURVE-FIT ANALYSIS ***

•FENCE DATA
 •FENCE 1,0. 1
 VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COFF 0.0 (2) Y COFF 0.1507E-01 (3) Z COFF 0.0 (4) CONSTANT 0.0
 ROOT MEAN SQUARE ERROR OF FIT= 0.0
 •FENCE NO. 2
 VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COFF 0.0 (2) Y COFF 0.212FE-01 (3) Z COFF 0.0 (4) CONSTANT 0.0
 KNOT MEAN SQUARE ERROR OF FIT= 0.0

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** RESULTS OF CURVE-FIT ANALYSIS ***

*REFLECTIVE SURFACE DATA

*REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD

	VALUES FOR COEFFICIENTS OF SURFACE				
(1)	X COEF	-0.2832E-02	(2)	Y COEF	0.0
(4)	XY COEF	0.0	(5)	XZ COEF	0.0
(7)	XSQ COEF	0.0	(8)	YSQ COEF	0.0
(10)	CONSTANT	-0.1000E 01			

ROOT MEAN SQUARE ERROR OF FIT= 0.9813E-17

*REFLECTIVE SURFACE NO. 2 IDENTIFICATION: TOP PANEL

	VALUES FOR COEFFICIENTS OF SURFACE				
(1)	X COEF	-0.9607E-18	(2)	Y COEF	0.0
(4)	XY COEF	0.0	(5)	XZ COEF	0.0
(7)	XSQ COEF	0.0	(8)	YSQ COEF	0.0
(10)	CONSTANT	-0.1000E 01			

ROOT MEAN SQUARE ERROR OF FIT= 0.2944E-16

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: LEFT FORWARD PANEL

	VALUES FOR COEFFICIENTS OF SURFACE				
(1)	X COEF	-0.4204E-16	(2)	Y COEF	-0.5556E-01
(4)	XY COEF	0.0	(5)	XZ COEF	0.0
(7)	XSQ COEF	0.0	(8)	YSQ COEF	0.0
(10)	CONSTANT	-0.1000E 01			

ROOT MEAN SQUARE ERROR OF FIT= 0.3A30E-15

*REFLECTIVE SURFACE NO. 4 IDENTIFICATION: RIGHT FORWARD PANEL

	VALUES FOR COEFFICIENTS OF SURFACE				
(1)	X COEF	-0.4204E-16	(2)	Y COEF	0.5556E-01
(4)	XY COEF	0.0	(5)	XZ COEF	0.0
(7)	XSQ COEF	0.0	(8)	YSQ COEF	0.0
(10)	CONSTANT	-0.1000E 01			

ROOT MEAN SQUARE ERROR OF FIT= 0.3A30E-15

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

*** RESULTS OF CURVE-FIT ANALYSIS ***

*REFLECTIVE SURFACE DATA

IDENTIFICATION: LEFT AFT PANEL
 REFLECTIVE SURFACE NO. 5
 VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.8547E-1A
 (4) XY COEF 0.0
 (7) XZ COEF 0.0
 (10) CONSTANT -0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.9047E-16

IDENTIFICATION: LEFT AFT PANEL
 REFLECTIVE SURFACE NO. 5
 VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.8547E-1A
 (4) XY COEF 0.0
 (7) XZ COEF 0.0
 (10) CONSTANT -0.1000E 01

*REFLECTIVE SURFACE NO. 6 IDENTIFICATION: RIGHT AFT PANEL

VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.8547E-1A
 (4) XY COEF 0.0
 (7) XZ COEF 0.0
 (10) CONSTANT -0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.9047E-16

VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.8547E-1A
 (4) XY COEF 0.0
 (7) XZ COEF 0.0
 (10) CONSTANT -0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.9047E-16

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 1

IDENTIFICATION: WINDSOFTFLD

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

A/C YAW

BOUNDARY POINTS -- BODY AXES
 (STATIONLINE, GUTTLINE, WATERLINE)
 X Y Z

	X	Y	Z	STATUS FLAGS FOR POINTS	TREFL	ISHAO	INTRF	CLINT	AZIMUTH	GLINT ELEVATION
-180.00	27.60	-1P.00	42.00	1	0	0	0	0	8.98	-P2+70
-180.00	54.00	-1R.00	72.00	1	0	0	0	0	8.97	-P2+68
-180.00	54.00	1R.00	72.00	1	0	0	0	0	8.79	-P2+68
-180.00	27.60	1P.00	42.00	1	0	0	0	0	8.81	-P2+70
-185.00	27.60	-1R.00	42.00	1	0	0	0	0	129.34	-44+51
-135.00	54.00	-1P.00	72.00	1	0	0	0	0	12R.22	-44+43
-135.00	54.00	1R.00	72.00	1	0	0	0	0	12P.20	-44+52
-135.00	27.60	1P.00	42.00	1	0	0	0	0	12R.32	-44+50
-90.00	27.60	-1R.00	42.00	0	1	0	0	0	-178.29	1.02
-90.00	54.00	-1R.00	72.00	0	1	0	0	0	-178.41	1.16
-90.00	54.00	1R.00	72.00	0	0	1	0	0	-178.41	1.17
-90.00	27.60	1P.00	42.00	0	0	1	0	0	-178.29	1.02
-45.00	27.60	-1R.00	42.00	0	0	0	0	0	-127.19	44.67
-45.00	54.00	-1R.00	72.00	0	0	0	0	0	-127.31	44.55
-45.00	54.00	1R.00	72.00	0	0	0	0	0	-127.29	44.64
-45.00	27.60	1P.00	42.00	0	0	0	0	0	-127.16	44.56
0.0	27.60	-1P.00	42.00	0	0	0	0	0	-0.09	R2+64
0.0	54.00	-1R.00	72.00	0	0	0	0	0	-0.09	R2+62
45.00	54.00	1R.00	72.00	0	0	0	0	0	0.09	R2+64
45.00	27.60	1P.00	42.00	0	0	0	0	0	0.09	R2+62
45.00	54.00	-1P.00	72.00	0	0	0	0	0	127.16	44.66
45.00	54.00	1R.00	72.00	0	0	0	0	0	127.29	44.64

45.00	18.00	42.00	0	0	0	127.19	44.47
90.00	27.60	-18.00	42.00	0	0	178.29	1.02
90.00	54.70	-18.00	72.00	0	0	178.41	1.17
90.00	54.90	18.00	72.00	0	1	178.41	1.16
90.00	27.60	18.00	42.00	0	1	178.29	1.02
97.50	27.60	-18.00	42.00	1	0	-128.32	-44.60
115.00	54.00	-18.00	72.00	1	0	-128.20	-44.52
135.00	54.70	18.00	72.00	1	0	-128.22	-44.43
135.00	27.60	18.00	42.00	1	0	-128.34	-44.51
180.00	27.60	-18.00	42.00	1	0	-8.81	-52.70
180.00	54.00	-18.00	72.00	1	0	-8.79	-52.68
180.00	54.70	18.00	72.00	1	0	-8.87	-52.68
180.00	27.60	18.00	42.00	1	0	-8.98	-52.70

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 2

IDENTIFICATION: TOP PANEL

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

XROT 150.00
 YROT 0.0
 ZROT 35.00
 DISTG 1000.00

A/C YAW	BOUNDARY POINTS -- PCOV AXES (STATIONLINE,BUTTLINE,WATERLINE)			STATUS	FLAGS FOR POINTS	AZIMUTH	GLINT ELEVATION
	X	Y	Z				
-180.00	66.00	-18.00	72.00	1	0	0	179.92 0.17
-180.00	118.80	-18.00	72.00	0	0	0	179.92 0.17
-180.00	118.80	18.00	72.00	0	0	0	-179.91 0.17
-180.00	66.00	18.00	72.00	0	0	0	-179.91 0.17
-135.00	66.00	-18.00	72.00	0	0	0	-179.78 0.17
-135.00	118.80	-18.00	72.00	0	0	0	-179.96 0.17
-135.00	118.80	18.00	72.00	0	0	0	-179.83 0.17
-135.00	66.00	18.00	72.00	0	0	0	-179.66 0.17
-90.00	66.00	-18.00	72.00	0	0	0	-179.60 0.17
-90.00	118.80	-18.00	72.00	0	0	0	-179.85 0.17
-90.00	118.80	18.00	72.00	0	0	0	-179.85 0.17
-90.00	66.00	18.00	72.00	0	0	0	-179.60 0.17
-45.00	66.00	-18.00	72.00	0	0	0	-179.66 0.17
-45.00	118.80	-18.00	72.00	0	0	0	-179.83 0.17
-45.00	118.80	18.00	72.00	0	0	0	-179.96 0.17
-45.00	66.00	18.00	72.00	0	0	0	-179.78 0.17
0.0	66.00	-18.00	72.00	0	0	0	-179.91 0.17
0.0	118.80	-18.00	72.00	0	0	0	179.91 0.17
0.0	118.80	18.00	72.00	0	0	0	179.78 0.17
45.00	66.00	-18.00	72.00	0	0	0	179.96 0.17
45.00	118.80	-18.00	72.00	0	0	0	179.83 0.17
45.00	118.80	18.00	72.00	0	0	0	179.83 0.17

45.00	66.00	18.00	72.00	0	0	0	179.66	0.17
90.00	66.70	-18.00	72.00	0	0	0	179.60	0.17
90.00	118.80	-18.00	72.00	0	0	0	179.85	0.17
90.00	119.80	18.00	72.00	0	0	0	179.85	0.17
90.00	66.90	18.00	72.00	0	0	0	179.60	0.17
135.00	66.70	-18.00	72.00	0	0	0	179.66	0.17
135.00	118.80	-18.00	72.00	0	0	0	179.83	0.17
135.00	118.80	18.00	72.00	0	0	0	179.96	0.17
135.00	66.00	18.00	72.00	0	0	0	179.78	0.17
180.00	66.00	-18.00	72.00	0	0	0	179.91	0.17
180.00	118.00	-18.00	72.00	0	0	0	179.91	0.17
180.00	118.80	18.00	72.00	0	0	0	-179.92	0.17
180.00	66.70	18.00	72.00	0	0	0	-179.92	0.17

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 3

IDENTIFICATION: LEFT FORWARD PANEL

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.0
A/C POLL ATTITUDE	0.0

XROT	150.00
YROT	0.0
ZROT	36.00
DISTG	1'000.00

A/C YAW	BOUNDARY POINTS -- BODY AXES (STATIONLINE,BOTTLINE,WATERLINE)			STATUS FLAGS FOR POINTS		
	X	Y	Z	IRFL	ISHAD	INTRF
-180.00	27.60	-18.00	42.00	1	0	0
-180.00	54.00	-18.00	72.00	1	0	0
-180.00	54.00	-18.00	42.00	1	0	0
-135.00	27.60	-18.00	42.00	1	0	0
-135.00	54.00	-18.00	72.00	1	0	0
-135.00	54.00	-18.00	42.00	1	0	0
-90.00	27.60	-18.00	42.00	1	1	0
-90.00	54.00	-18.00	72.00	1	1	0
-90.00	54.00	-18.00	42.00	1	0	0
-45.00	27.60	-18.00	42.00	1	0	0
-45.00	54.00	-18.00	72.00	1	0	0
-45.00	54.00	-18.00	42.00	1	0	0
0.0	27.60	-18.00	42.00	0	0	0
0.0	54.00	-18.00	72.00	0	0	0
0.0	54.00	-18.00	42.00	0	0	0
45.00	27.60	-18.00	42.00	0	0	0
45.00	54.00	-18.00	72.00	0	0	0
45.00	54.00	-18.00	42.00	0	0	0
90.00	27.60	-18.00	42.00	0	0	0
90.00	54.00	-18.00	72.00	0	0	0
90.00	54.00	-18.00	42.00	0	0	0
135.00	27.60	-18.00	42.00	0	0	0
135.00	54.00	-18.00	72.00	0	0	0
135.00	54.00	-18.00	42.00	0	0	0

135.00	54.00	-18.00	42.00	0	0	0	90.26
186.00	27.00	-18.00	42.00	0	0	0	177.92
180.00	54.00	-18.00	72.00	0	0	0	177.92
180.00	54.00	-18.00	42.00	0	0	0	177.92

FLAT SURFACES WITH FFNCES

SAMPLE CASE ONE

*** SUN CLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 4

IDENTIFICATION: RIGHT FORWARD FFNC

SUN ELEVATION: 3.0

A/C PITCH ATTITUDE 0.0

A/C ROLL ATTITUDE 0.0

XROT 150.00

YROT 0.0

ZROT 36.00

DISTG 1000.00

A/C YAW
FOUNDRY POINTS -- EASY AXES
(STATIONLINE*PUTLINE*WATERLINE)
X Y Z

A/C YAW	X	Y	Z	STATUS FLAGS FOR POINTS	CLINT	CLINT	AZIMUTH ELEVATION
-180.00	27.60	18.00	42.00	0	0	0	-177.92 0.03
-180.00	54.00	18.00	72.00	0	0	0	-177.92 0.17
-180.00	54.00	18.00	42.00	0	0	0	-177.92 0.03
-135.00	27.60	18.00	42.00	0	0	0	-60.35 0.05
-135.00	54.00	18.00	72.00	0	0	0	-90.26 0.17
-135.00	54.00	18.00	42.00	0	0	0	-90.26 0.03
-90.00	27.60	18.00	42.00	0	0	0	1.41 0.03
-90.00	54.00	18.00	72.00	0	0	0	1.54 0.17
-90.00	54.00	18.00	42.00	0	0	0	1.54 0.03
-45.00	27.60	18.00	42.00	0	0	0	89.53 0.03
-45.00	54.00	18.00	72.00	0	0	0	89.62 0.17
-45.00	54.00	18.00	42.00	0	0	0	89.62 0.03
0.0	27.60	18.00	42.00	0	0	0	179.91 0.03
0.0	54.00	18.00	72.00	0	0	0	179.91 0.17
0.0	54.00	18.00	42.00	0	0	0	179.91 0.03
45.00	27.60	18.00	42.00	1	0	0	-89.65 0.03
45.00	54.00	18.00	72.00	1	0	0	-89.74 0.17
45.00	54.00	18.00	42.00	1	0	0	-89.74 0.03
90.00	27.60	18.00	42.00	1	1	0	-1.42 0.03
90.00	54.00	18.00	72.00	1	1	0	-1.54 0.17
90.00	54.00	18.00	42.00	1	0	0	-1.54 0.03
135.00	27.60	18.00	42.00	1	0	0	90.47 0.03
135.00	54.00	18.00	72.00	1	0	0	90.38 0.17

135.00	18.00	42.00	90.00	0.03
182.00	18.00	42.00	178.10	0.03
160.00	18.00	72.00	178.09	0.17
180.00	18.00	42.00	178.09	0.13
54.00	18.00	42.00	0	0
27.00	18.00	42.00	0	0
54.00	18.00	72.00	0	0
135.00	18.00	42.00	0	0
182.00	18.00	42.00	0	0
160.00	18.00	72.00	0	0
180.00	18.00	42.00	0	0

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

REFLECTIVE SURFACE NO. 6

*** SUN GLINT SIGNATURE ***

SUN ELEVATION 0° 0
 A/C PITCH ATTITUDE 0° 0
 A/C ROLL ATTITUDE 0° 0

IDENTIFICATION: LEFT EFT PANFL

XROT 160.00
 YROT 0.0
 ZROT 36.00
 ObjTG 1000.00

A/C YAW	BOUNDARY POINTS -- PWDY AXFS (STATIONLINE,BUTTLINE,WAPELINE)			STATUS FLAGS FOR POINTS THRF1 THRF2 ISHAD			AZIMUTH ELEVATION	GLINT ELEVATION
	X	Y	Z					
-180.00	66.00	-18.00	72.00		1	0	0	-178.09 0.17
-180.00	118.00	-18.00	72.00		1	0	0	-178.09 0.17
-180.00	118.00	-18.00	42.00		1	0	0	-178.09 0.03
-180.00	66.00	-18.00	42.00		1	0	0	-178.09 0.03
-135.00	66.00	-18.00	72.00		1	0	0	-90.34 0.17
-135.00	118.00	-18.00	72.00		1	0	0	-90.17 0.17
-135.00	118.00	-18.00	42.00		1	0	0	-90.17 0.03
-135.00	66.00	-18.00	42.00		1	0	0	-90.34 0.03
-90.00	66.00	-18.00	72.00		1	0	0	1.60 0.17
-90.00	118.00	-18.00	72.00		1	0	0	1.85 0.17
-90.00	118.00	-18.00	42.00		1	0	0	1.85 0.03
-90.00	66.00	-18.00	42.00		1	0	0	1.60 0.03
-45.00	66.00	-18.00	72.00		1	0	0	89.78 0.17
-45.00	118.00	-18.00	72.00		1	0	0	89.96 0.17
-45.00	118.00	-18.00	42.00		1	0	0	89.96 0.03
-45.00	66.00	-18.00	42.00		1	0	0	89.78 0.03
0.0	66.00	-18.00	72.00		0	0	0	-179.91 0.17
0.0	118.00	-18.00	72.00		0	0	0	-179.91 0.03
0.0	118.00	-18.00	42.00		0	0	0	-179.91 0.03
45.00	66.00	-18.00	42.00		0	0	0	-89.66 0.17
45.00	118.00	-18.00	42.00		0	0	0	-89.83 0.17
45.00	118.00	-18.00	42.00		0	0	0	-89.83 0.03

45.00	66.00	-18.00	42.00	0	0	0	-19.66	0.03
90.00	66.00	-18.00	72.00	0	0	0	-1.60	0.17
90.00	118.30	-18.00	72.00	0	0	0	-1.85	0.17
90.00	118.40	-18.00	42.00	0	0	0	-1.85	0.03
90.00	66.00	-18.00	42.00	0	0	0	-1.60	0.03
135.00	66.00	-18.00	72.00	0	0	0	90.22	0.17
135.00	118.40	-18.00	72.00	0	0	0	93.04	0.17
135.00	118.40	-18.00	42.00	0	0	0	90.04	0.03
135.00	66.00	-18.00	42.00	0	0	0	90.22	0.03
180.00	66.00	-18.00	72.00	0	0	0	177.92	0.17
180.00	118.40	-18.00	72.00	0	0	0	177.92	0.17
180.00	118.40	-18.00	42.00	0	0	0	177.92	0.03
180.00	66.00	-18.00	42.00	0	0	0	177.92	0.03

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

REFLECTIVE SURFACE NO. 6 IDENTIFICATION: RIGHT AFT FANFL

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE C.0
 A/C ROLL ATTITUDE 0.0

*** SUN GLINT SIGNATURE ***

A/C YAW	FOUNDRY POINTS -- ROCK AXES			STATUS FLAGS FOR POINTS INRFL INSHAD INTRF	CLINIC	AZIMUTH	ELLEVATION
	X	Y	Z				
-180.00	66.00	18.00	72.00	0	0	0	0.17
-180.00	118.80	18.00	72.00	0	0	0	-177.92
-180.00	118.80	18.00	42.00	0	0	0	-177.92
-180.00	66.00	18.00	42.00	0	0	0	-177.92
-135.00	66.00	18.00	72.00	0	0	0	-90.22
-135.00	118.80	18.00	72.00	0	0	0	-90.22
-135.00	118.80	18.00	42.00	0	0	0	-90.22
-135.00	66.00	18.00	42.00	0	0	0	-90.22
-90.00	66.00	18.00	72.00	0	0	0	1.60
-90.00	118.80	18.00	72.00	0	0	0	1.65
-90.00	118.80	18.00	42.00	0	0	0	1.65
-90.00	66.00	18.00	42.00	0	0	0	1.60
-45.00	66.00	18.00	72.00	0	0	0	0.03
-45.00	118.80	18.00	72.00	0	0	0	0.03
-45.00	118.80	18.00	42.00	0	0	0	0.03
-45.00	66.00	18.00	42.00	0	0	0	0.03
0.0	66.00	18.00	72.00	0	0	0	0.03
0.0	118.80	18.00	72.00	0	0	0	0.03
0.0	118.80	18.00	42.00	0	0	0	0.03
45.00	66.00	18.00	42.00	0	0	0	0.03
45.00	118.80	18.00	42.00	0	0	0	0.03
45.00	66.00	18.00	72.00	1	0	0	179.91
45.00	118.80	18.00	72.00	1	0	0	179.91
45.00	66.00	18.00	42.00	0	0	0	-89.96
45.00	118.80	18.00	42.00	0	0	0	-89.96

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

SUN GLINT SIGNATURE

SUN ELEVATION 0.0
A/C F/TCH ATTITUDE 2.0
A/C ROLL ATTITUDE 0.0

OBSERVER ANGLE
DEGREES

10.000

OBSERVER ANGLE
DEGREES

10.000

-5.000

-5.000

0.0

0.0

60

60

-10.000

-10.000

-10.000

-10.000

-180.000 -140.000 -100.000 -60.000 -20.000 20.000 60.000 100.000 140.000 180.000

GLINT AZIMUTH WITH RESPECT TO SUN - DEGREES

PROBABILITY: 0.064

* KEY TO PLOT SYMBOLS

SYMBOL	A/C HEADING (DEG)	SYMBOL	A/C HEADING (DEG)
A	-180.00	F	95.00
P	-135.00	G	90.00
C	-90.00	R	175.00
D	-45.00	M	180.00
F	0.0		

FLAT SURFACE WITH FENCES SAMPLE CASE ONE

*** SUP CLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD

SUN ELEVATION 30.00
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

MNT 150.00
 YAC 0.0
 ZPT 36.00
 DISTG 1000.00

A/C YAW	BOUNDARY POINTS -- POOL AXES (STATIONLINE,BUTLINE,WATERLINE)			STATUS FLARS FOR POINTS INPL ISHAD INTRE			AZIMUTH ELEVATION	CLINT
	X	Y	Z	INPL	ISHAD	INTRE		
-180.00	27.60	-18.00	42.00	1	0	0	178.69	-67.07
-180.00	54.00	-18.00	72.00	1	0	0	178.69	-67.09
-180.00	54.00	18.00	72.00	1	0	0	178.66	-67.10
-180.00	27.60	18.00	42.00	1	0	0	178.67	-67.07
-135.00	27.60	-18.00	42.00	1	0	0	169.76	-41.93
-135.00	54.00	-18.00	72.00	1	0	0	169.65	-41.98
-135.00	54.00	18.00	72.00	1	0	0	169.75	-41.95
-135.00	27.60	18.00	42.00	1	0	0	169.85	-42.00
-90.00	27.60	-18.00	42.00	0	0	0	-148.64	-2.74
-90.00	54.00	-18.00	72.00	0	0	0	-148.75	-2.60
-90.00	54.00	18.00	72.00	0	0	0	-148.66	-2.61
-90.00	27.60	18.00	42.00	0	0	0	-148.55	-2.75
-45.00	27.60	-18.00	42.00	0	0	0	-91.49	32.75
-45.00	54.00	-18.00	72.00	0	0	0	-91.59	32.99
-45.00	54.00	18.00	72.00	0	0	0	-91.47	32.95
-45.00	27.60	18.00	42.00	0	0	0	-91.38	32.61
0.0	27.60	-18.00	42.00	0	0	0	-0.09	52.42
0.0	54.00	-18.00	72.00	0	0	0	-0.09	52.54
0.0	54.00	18.00	72.00	0	0	0	0.09	52.54
0.0	27.60	18.00	42.00	0	0	0	0.09	52.42
45.00	27.60	-18.00	42.00	0	0	0	91.38	32.81
45.00	54.00	-18.00	72.00	0	0	0	91.47	32.95
45.00	54.00	18.00	72.00	0	0	0	91.59	32.89

45.00	27.60	18.00	42.00	0	0	0	91.49	32.75
90.00	27.60	-18.00	42.00	0	0	0	148.55	-2.75
90.00	54.00	-18.00	72.00	0	0	0	148.66	-2.61
90.00	54.00	18.00	72.00	0	0	0	148.75	-2.60
90.00	27.60	18.00	42.00	0	0	0	148.64	-2.74
135.00	27.60	-18.00	42.00	1	0	0	-169.85	-42.00
135.00	54.00	-18.00	72.00	1	0	0	-169.75	-41.95
135.00	54.00	18.00	72.00	1	0	0	-169.65	-41.98
135.00	27.60	18.00	42.00	1	0	0	-169.76	-41.93
180.00	27.60	-18.00	42.00	1	0	0	-178.87	-67.07
180.00	54.00	-18.00	72.00	1	0	0	-178.86	-67.10
180.00	54.00	18.00	72.00	1	0	0	-178.69	-67.09
180.00	27.60	18.00	42.00	1	0	0	-178.69	-67.07

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 2

IDENTIFICATION: TOP PANEL

SUN ELEVATION 30.00
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

XROT 150.00
 YROT 0.0
 ZROT 36.00
 DISTG 1000.00

A/C YAW	POINTER POINTS -- PCNY AXES (STATIONLINE, RUTTLINE, WATERLINE)			STATUS	FLAGS FOR POINTS TIRFL	POINTS 1SHAD	GLINT AZIMUTH	ELEVATION
	X	Y	Z					
-180.00	66.00	-18.00	72.00	0	0	0	179.92	29.95
-180.00	118.00	-18.00	72.00	0	0	0	179.92	30.06
-180.00	118.00	18.00	72.00	0	0	0	-179.91	30.06
-180.00	66.00	18.00	72.00	0	0	0	-179.91	29.96
-135.00	66.00	-18.00	72.00	0	0	0	-179.78	29.98
-135.00	118.00	-18.00	72.00	0	0	0	-179.96	30.06
-135.00	118.00	18.00	72.00	0	0	0	-179.83	30.11
-135.00	66.00	18.00	72.00	0	0	0	-179.66	30.03
-90.00	66.00	-18.00	72.00	0	0	0	-179.60	30.09
-90.00	118.00	-18.00	72.00	0	0	0	-179.85	30.17
-90.00	118.00	18.00	72.00	0	0	0	-179.85	30.17
-45.00	66.00	-18.00	72.00	0	0	0	-179.66	30.22
-45.00	118.00	-18.00	72.00	0	0	0	-179.83	30.15
-45.00	118.00	18.00	72.00	0	0	0	-179.96	30.20
-45.00	66.00	18.00	72.00	0	0	0	-179.78	30.28
0.0	66.00	-18.00	72.00	0	0	0	-179.91	30.30
0.0	118.00	-18.00	72.00	0	0	0	-179.91	30.19
0.0	118.00	18.00	72.00	0	0	0	179.91	30.19
45.00	66.00	-18.00	72.00	0	0	0	179.91	30.10
45.00	118.00	-18.00	72.00	0	0	0	179.78	30.28
45.00	118.00	18.00	72.00	0	0	0	179.96	30.20
45.00	66.00	18.00	72.00	0	0	0	179.83	30.15

45.00	66.00	18.00	72.00	0	0	179.66	30.22
90.00	66.00	-18.00	72.00	0	0	179.60	30.17
90.00	118.80	-18.00	72.00	0	0	179.85	30.17
90.00	118.80	18.20	72.00	0	0	179.85	30.09
90.00	66.22	18.00	72.00	0	0	179.60	30.09
135.00	66.00	-18.00	72.00	0	0	179.66	30.03
135.00	118.80	-18.00	72.00	0	0	179.83	30.11
135.00	118.80	18.00	72.00	0	0	179.96	30.06
135.00	66.00	18.00	72.00	0	0	179.78	29.98
180.00	66.00	-18.00	72.00	0	0	179.91	29.96
180.00	118.80	-18.00	72.00	0	0	179.91	30.06
180.00	118.80	18.00	72.00	0	0	-179.92	30.06
180.00	66.00	18.00	72.00	0	0	-179.92	29.95

FLAT SURFACE WITH REFLECTIONS

SAMPLE CASE ONE

*** SIMULATED SURFACE ***

REFLECTIVE SURFACE NO. *

SIM ELEVATION 30.00
A/C PITCH ATTITUDE 0.0
A/C ROLL ATTITUDE 0.0

IDENTIFICATION: LEFT FORWARD PANEL

XROT 150.00
YROT 0.0
ZROT 36.00
DISTC 100.00

A/C YAW

FOUNDARY POINTS -- ECYC AXES
(STATIONLINE, BOTTOMLINE, WATERLINE)

GLINT AZIMUTH ELEVATION,

			STATUS	FLAGS FOR POINTS	
			INFL	ISAD	TWTF
-160.00	27.60	-18.00	42.00	1	0
-140.00	54.00	-18.00	72.00	1	0
-120.00	54.00	-18.00	42.00	1	0
-105.00	27.60	-18.00	42.00	1	0
-115.00	54.00	-18.00	72.00	1	0
-135.00	54.00	-18.00	42.00	1	0
-90.00	27.60	-18.00	42.00	1	0
-90.00	54.00	-18.00	72.00	1	0
-90.00	54.00	-18.00	42.00	1	0
-45.00	27.60	-18.00	42.00	1	0
-45.00	54.00	-18.00	72.00	1	0
-45.00	54.00	-18.00	42.00	1	0
0.0	27.60	-18.00	42.00	0	0
0.0	54.00	-18.00	72.00	0	0
0.0	54.00	-18.00	42.00	0	0
45.00	27.60	-18.00	42.00	0	0
45.00	54.00	-18.00	72.00	0	0
45.00	54.00	-18.00	42.00	0	0
90.00	27.60	-18.00	42.00	0	0
90.00	54.00	-18.00	72.00	0	0
90.00	54.00	-18.00	42.00	0	0
135.00	27.60	-18.00	42.00	0	0
135.00	54.00	-18.00	72.00	0	0
135.00	54.00	-18.00	42.00	0	0

135.00						
54.70	-18.00	42.00		0	0	90.26 -29.81
160.00	27.60	-18.00	42.00	0	0	177.92 -29.72
180.00	54.70	-18.00	72.00	0	0	177.92 -29.77
180.00	54.00	-18.00	42.00	0	0	177.02 -29.78

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BOEING VERTOL CO PHILADELPHIA PA
CANOPY SUN GLINT EVALUATION COMPUTER PROGRAM.(U)
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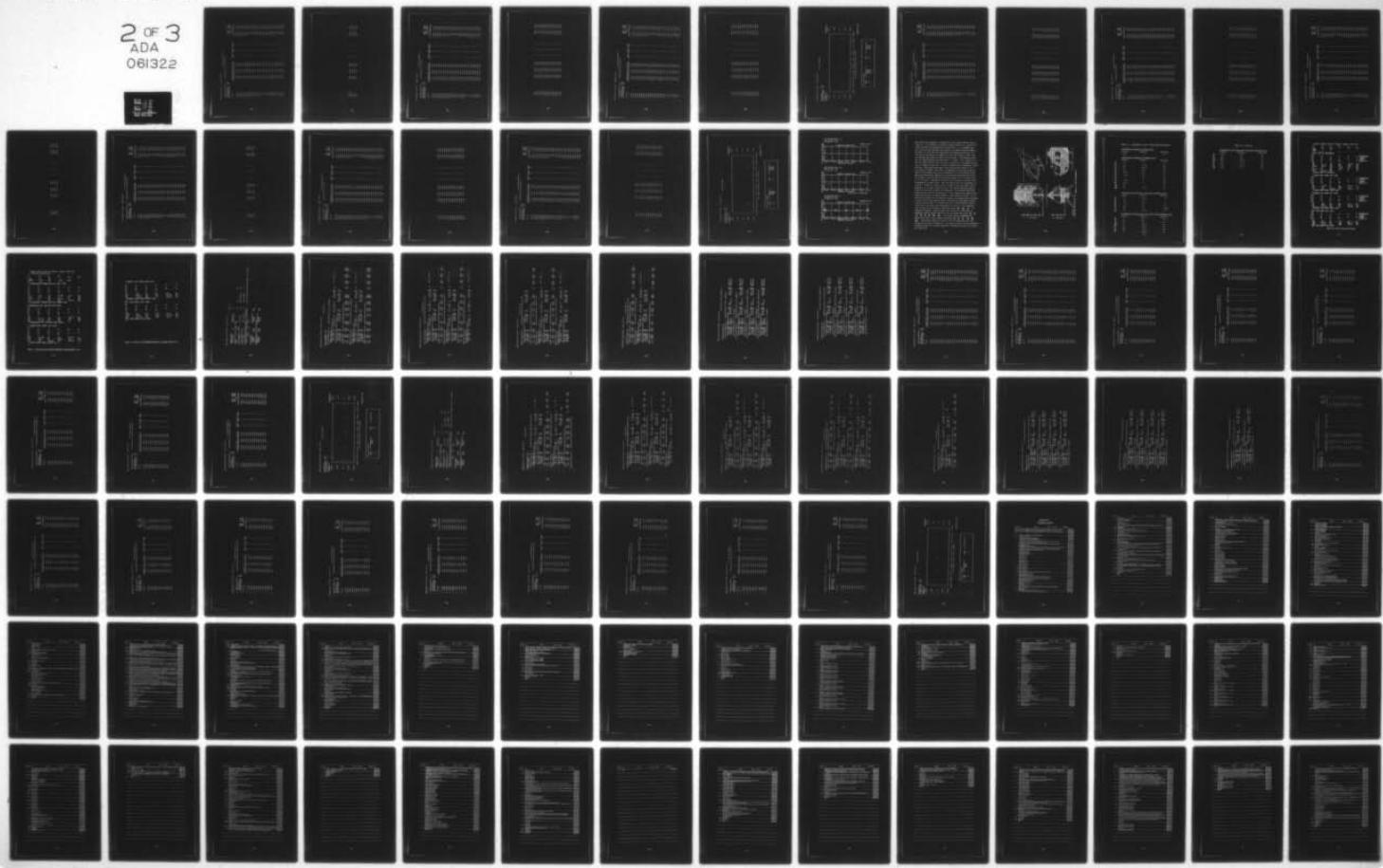
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FLAT SURFACE: WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

EFFECTIVE SURFACE NO. 4

SUN ELEVATION 30.0°
 A/C PITCH ATTITUDE 5.0°
 A/CROLL ATTITUDE 0.0°

IDENTIFICATION: RIGHT FORWARD FENCE

XROT 150.00
 YROT 0.0
 ZROT 36.00
 DIST 1210.00

BOUNDARY POINTS - X AND Y AXES
 (STATION LINE + RUTLINE + WATERLINE)

GLINT AZIMUTH ELEVATION

A/C YAW	X	Y	Z	INFRLL	ISHADE	INFTRR	POINT STATUS
-180.00	27.60	1P.00	42.00	0	0	0	-177.92
-165.00	54.00	1P.00	72.00	0	0	0	-177.92
-150.00	54.00	1P.00	42.00	0	0	0	-177.92
-135.00	27.60	1P.00	42.00	0	0	0	-177.92
-135.00	54.00	1P.00	72.00	0	0	0	-177.92
-135.00	54.00	1P.00	42.00	0	0	0	-177.92
-90.00	27.60	1P.00	42.00	0	0	0	-177.92
-90.00	54.00	1P.00	72.00	0	0	0	-177.92
-90.00	54.00	1P.00	42.00	0	0	0	-177.92
-45.00	27.60	1P.00	42.00	0	0	0	-177.92
-45.00	54.00	1P.00	72.00	0	0	0	-177.92
-45.00	54.00	1P.00	42.00	0	0	0	-177.92
0.0	27.60	1P.00	42.00	0	0	0	-177.92
0.0	54.00	1P.00	72.00	0	0	0	-177.92
0.0	54.00	1P.00	42.00	0	0	0	-177.92
45.00	27.60	1P.00	42.00	1	0	0	-177.92
45.00	54.00	1P.00	72.00	1	0	0	-177.92
45.00	54.00	1P.00	42.00	1	0	0	-177.92
90.00	27.60	1P.00	42.00	1	0	0	-177.92
90.00	54.00	1P.00	72.00	1	0	0	-177.92
90.00	54.00	1P.00	42.00	1	0	0	-177.92
135.00	27.60	1P.00	42.00	1	0	0	-177.92
135.00	54.00	1P.00	72.00	1	0	0	-177.92

135.00	54.00	18.00	42.00	1	9	0	0	07.38	-29.86
180.90	27.60	18.00	42.00	1	3	0	0	178.10	-29.73
180.00	54.00	18.00	72.00	1	0	0	0	178.09	-29.67
180.00	54.00	18.00	42.00	1	0	0	0	178.09	-29.78

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

*** SUM GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 5

SUN ELEVATION 30.39
 A/C PITCH ATTITUDE 2.0
 A/C ROLL ATTITUDE 2.0

IDENTIFICATION: LEFT AFT PANEL

XROT 150.00
 YROT 0.0
 ZROT 36.00
 DISTG 1000.00

A/C YAW
 BOUNDARY POINTS -- PONY AXES
 (STATIONLINE,BUTLINE,WATERLINE)
 X Y

			STATUS FLAGS FOR POINTS	GLINT
			INFL ISHAO INTRF	AZIMUTH ELEVATION
-180.00	66.00	-18.00	72.00	1 0 0 -178.09 -79.78
-180.00	118.60	-18.00	72.00	1 0 0 -178.09 -79.81
-180.00	118.60	-18.00	42.00	1 0 0 -178.09 -79.91
-180.00	66.00	-18.00	42.00	1 0 0 -178.09 -79.81
-180.00	66.00	-18.00	72.00	1 0 0 -90.34 -79.77
-180.00	118.60	-18.00	72.00	1 0 0 -90.17 -79.85
-180.00	118.60	-18.00	42.00	1 0 0 -90.17 -79.96
-180.00	66.00	-18.00	42.00	1 0 0 -90.34 -79.88
-90.00	66.00	-18.00	72.00	1 0 0 1.60 -29.91
-90.00	118.60	-18.00	72.00	1 0 0 1.65 -29.91
-90.00	118.60	-18.00	42.00	1 0 0 1.65 -30.02
-90.00	66.00	-18.00	42.00	1 0 0 1.60 -30.02
-45.00	66.00	-18.00	72.00	1 0 0 89.78 -30.02
-45.00	118.60	-18.00	72.00	1 0 0 89.06 -29.94
-45.00	118.60	-18.00	42.00	1 0 0 89.96 -30.05
-45.00	66.00	-18.00	42.00	1 0 0 89.78 -30.13
0.0	66.00	-18.00	72.00	0 0 0 -179.91 -30.04
0.0	118.60	-18.00	72.00	0 0 0 -179.91 -30.15
45.00	66.00	-18.00	72.00	0 0 0 -89.66 -29.97
45.00	118.60	-18.00	72.00	0 0 0 -99.83 -29.89
45.00	118.60	-18.00	42.00	0 0 0 -89.83 -30.05

65.00	66.00	-18.00	42.00	0	0	-89.66	-30.07
90.00	66.00	-18.00	72.00	0	0	-1.60	-29.64
90.00	118.80	-18.00	72.00	0	0	-1.85	-29.83
90.00	118.80	-18.00	42.00	0	0	-1.85	-29.94
90.00	66.00	-18.00	42.00	0	0	-1.60	-29.94
135.00	66.00	-18.00	72.00	0	0	90.22	-29.72
135.00	118.00	-18.00	72.00	0	0	90.04	-29.80
135.00	118.00	-18.00	42.00	0	0	90.04	-29.91
135.00	66.00	-18.00	42.00	0	0	90.22	-29.83
160.00	46.00	-18.00	72.00	0	0	177.92	-29.70
160.00	118.00	-18.00	72.00	0	0	177.92	-29.81
160.00	118.80	-18.00	42.00	0	0	177.92	-29.91
160.00	66.00	-18.00	42.00	0	0	177.92	-29.80

FLAT SURFACE & TIM FENCES

SAMPLE CASE ONE

*** SUM GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 6

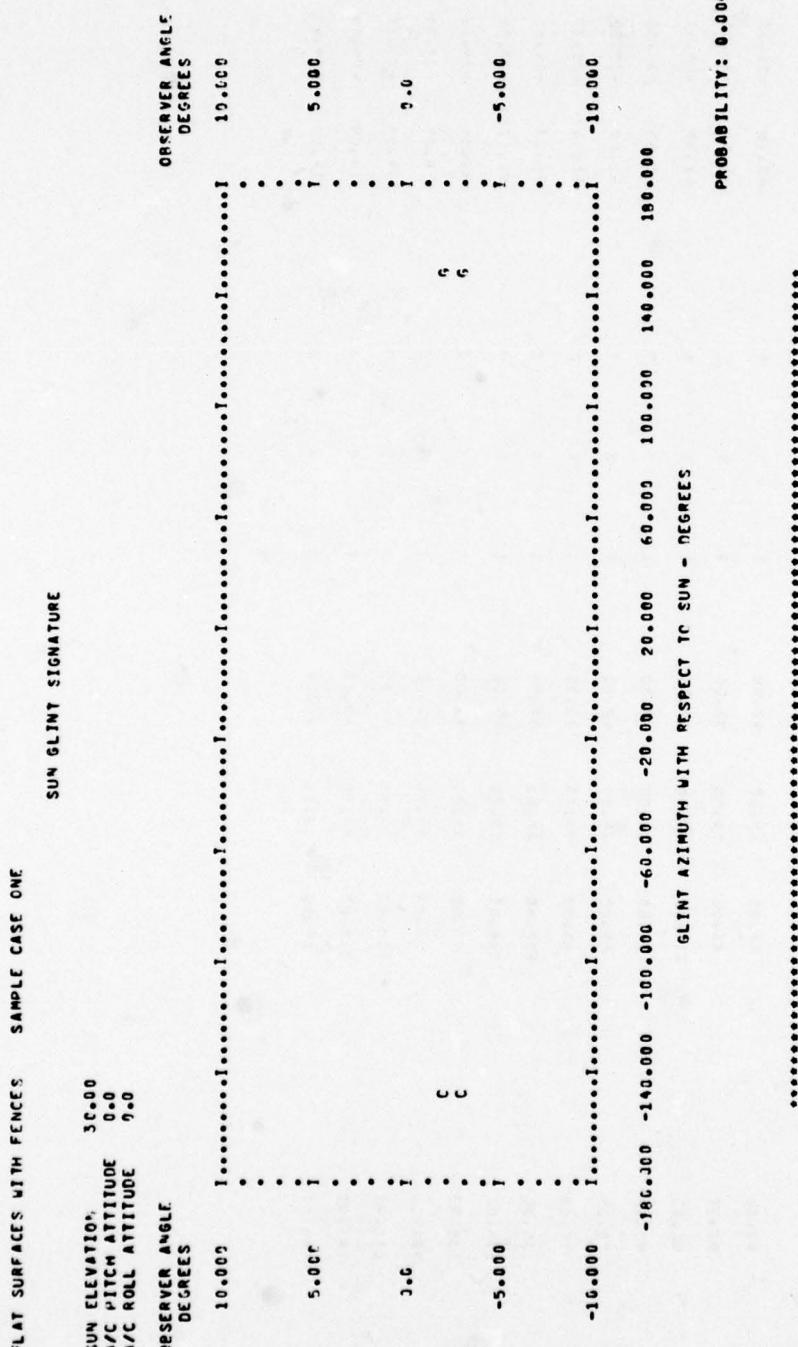
IDENTIFICATION: RIGHT AFT PANEL

SUN ELEVATION 30.00
 A/C PITCH ATTITUDE 0.2
 A/C ROLL ATTITUDE 0.0

XROT 150.00
 YROT 0.0
 ZROT 36.00
 DIST 100.00

A/C YAW	BOUNDARY POINTS -- BODY AXES (STATIONLINE, RULLINE, WATERLINE)	STATUS FLAGS FOR POINTS INRFL ISMAD INTPF	GLINT AZIMUTH ELEVATION
-180.00	66.00 18.00 72.00	0 0 0	-177.92 -29.70
-160.00	118.80 18.00 72.00	0 0 0	-177.92 -29.81
-140.00	118.80 18.00 42.00	0 0 0	-177.92 -29.91
-120.00	66.00 18.00 42.00	0 0 0	-177.92 -29.80
-100.00	66.00 18.00 72.00	0 0 0	-90.22 -29.72
-80.00	118.80 18.00 72.00	0 0 0	-90.04 -29.80
-60.00	118.80 18.00 42.00	0 0 0	-90.04 -29.91
-40.00	66.00 18.00 42.00	0 0 0	-90.22 -29.83
-20.00	66.00 18.00 72.00	0 0 0	1.60 -29.84
-90.00	118.80 18.00 72.00	0 0 0	1.05 -29.53
-90.00	118.80 18.00 42.00	0 0 0	1.05 -29.94
-90.00	66.00 18.00 42.00	0 0 0	1.60 -29.94
-45.00	66.00 18.00 72.00	0 0 0	89.66 -29.97
-45.00	118.80 18.00 72.00	0 0 0	89.83 -29.59
-45.00	118.80 18.00 42.00	0 0 0	89.83 -29.00
-45.00	66.00 18.00 42.00	0 0 0	179.91 -30.04
0.0	66.00 18.00 72.00	0 0 0	179.91 -30.15
0.0	118.80 18.00 72.00	0 0 0	179.91 -30.02
0.0	118.80 18.00 42.00	0 0 0	179.91 -30.06
45.00	66.00 18.00 72.00	1 0 0	-99.96 -29.94
45.00	118.80 18.00 72.00	1 0 0	-99.96 -29.94
45.00	118.80 18.00 42.00	1 0 0	-99.96 -29.94

45.00	66.00	18.00	42.00	1	0	0	-89.78	-35.13
90.39	66.00	18.00	72.00	1	0	0	-1.60	-29.91
90.00	118.00	18.00	72.00	1	0	0	-1.85	-29.51
90.00	118.00	18.00	42.00	1	0	0	-1.85	-30.02
90.00	66.00	18.00	42.00	1	0	0	-1.60	-35.02
66.00	66.00	18.00	72.00	1	0	0	90.34	-29.77
135.00	118.00	18.00	72.00	1	0	0	90.17	-29.85
135.00	118.00	18.00	42.00	1	0	0	90.17	-29.96
135.00	66.00	18.00	42.00	1	0	0	90.34	-29.88
135.00	66.00	18.00	72.00	1	0	0	178.09	-29.70
180.00	118.00	18.00	72.00	1	0	0	178.09	-29.81
180.00	118.00	18.00	42.00	1	0	0	178.09	-29.91
180.00	66.00	18.00	42.00	1	0	0	178.09	-29.81



FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSFIELD

SUN ELEVATION	60.00
A/C PITCH ATTITUDE	0.0
A/C ROLL ATTITUDE	2.0

XROT	150.00
YROT	0.0
ZROT	36.00
NTSG	1700.00

A/C YAW	BOUNDARY POINTS -- BODY AXES (STATIONLINE, RUTLINE, WATERLINE)			STATUS FLAGS FOR POINTS INFL	GLINT AZIMUTH	ELEVATION
	X	Y	Z			
-180.00	27.60	-18.00	42.00	0 0 0	-179.71	-37.00
-180.50	54.00	-18.00	72.00	0 0 0	-179.71	-36.97
-180.00	54.00	18.00	72.00	0 0 0	-179.54	-36.97
-180.00	27.60	18.00	42.00	0 0 0	-179.54	-37.00
-135.00	27.60	-18.00	42.00	0 0 0	-178.32	-27.18
-135.00	54.00	-18.00	72.00	0 0 0	-178.37	-27.12
-135.00	54.00	18.00	72.00	0 0 0	-178.21	-27.14
-135.60	27.60	18.00	42.00	0 0 0	-178.16	-27.21
-90.00	27.60	-18.00	42.00	0 0 0	-118.95	-5.74
-90.00	54.00	-18.00	72.00	0 0 0	-119.01	-5.61
-90.00	54.00	18.00	72.00	0 0 0	-118.86	-5.62
-90.00	27.60	18.00	42.00	0 0 0	-118.80	-5.75
-45.00	27.60	-18.00	42.00	0 0 0	-66.23	13.81
-45.00	54.00	-18.00	72.00	0 0 0	-66.27	13.98
-45.00	54.00	18.00	72.00	0 0 0	-66.11	13.99
-45.00	27.60	18.00	42.00	0 0 0	-66.07	13.83
0.0	27.60	-18.00	42.00	0 0 0	-0.99	22.51
0.0	54.00	-18.00	72.00	0 0 0	-0.09	22.68
0.0	54.00	18.00	72.00	0 0 0	0.09	22.68
0.0	27.60	18.00	42.00	0 0 0	0.09	22.51
45.00	27.60	-18.00	42.00	0 0 0	66.07	13.83
45.00	54.00	-18.00	72.00	0 0 0	66.11	13.99
45.00	54.00	18.00	72.00	0 0 0	66.27	13.98

45.00	27.60	1A.00	42.00	0	0	0	66.23	13.61
90.00	27.60	-1A.00	42.00	0	0	0	118.80	-5.75
90.00	54.00	-1P.00	72.00	0	0	0	11A.86	-5.62
93.00	54.00	1A.00	72.00	0	0	0	119.31	-5.61
96.00	27.60	1A.00	42.00	0	0	0	118.95	-5.74
135.00	27.60	-1A.00	42.00	0	0	0	158.16	-27.21
135.00	54.00	-1A.00	72.00	0	0	0	158.21	-27.14
135.00	54.00	1A.00	72.00	0	0	0	158.37	-27.12
135.00	27.60	1A.C0	42.00	0	0	0	15A.32	-27.18
180.00	27.60	-1A.00	42.00	0	0	0	179.54	-27.20
180.00	54.00	1A.00	72.00	0	0	0	179.58	-26.97
180.00	54.00	-1P.00	72.00	0	0	0	179.71	-26.97
180.00	27.60	1A.00	42.00	0	0	0	179.71	-27.00

FLAT SURFACES WITH FENCES SAMPLE CASE CME

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 2

IDENTIFICATION: TOP PANEL

SUN ELEVATION 60.000
 A/C F/TCH ATTITUDE 0.0
 A/C R/G/L ATTITUDE 0.0

XROT 150.00
 YROT 0.0
 ZROT 20.00
 NISTG 1000.00

A/C YAW	BOUNDARY POINTS -- BODY AXES			STATUS FLAGS FOR POINTS			GLINT AZIMUTH	GLINT ELEVATION
	X	Y	Z	1STATIONLINE,BUTTLINE,VALVEPLTF, TRPLF	ISMAN	INPF		
-180.00	66.00	-1P.00	72.00	0	0	0	179.92	59.87
-180.00	11A.00	-1P.00	72.00	0	0	0	179.92	59.87
-180.00	11A.00	1P.00	72.00	0	0	0	-179.91	59.98
-180.00	66.00	1P.00	72.00	0	0	0	-179.91	59.87
-180.00	66.00	-1P.00	72.00	0	0	0	-179.76	59.89
-180.00	11A.00	-1P.00	72.00	0	0	0	-179.06	59.97
-180.00	11A.00	1P.00	72.00	0	0	0	-179.83	60.02
-180.00	66.00	1P.00	72.00	0	0	0	-179.66	59.95
-180.00	66.00	-1P.00	72.00	0	0	0	-179.69	60.01
-90.00	11A.00	-1P.00	72.00	0	0	0	-179.85	60.01
-90.00	11A.00	1P.00	72.00	0	0	0	-179.85	60.08
-90.00	66.00	1P.00	72.00	0	0	0	-179.60	60.08
-90.00	66.00	-1P.00	72.00	0	0	0	-179.66	60.14
-45.00	66.00	-1P.00	72.00	0	0	0	-179.83	60.06
-45.00	11A.00	-1P.00	72.00	0	0	0	-179.91	50.22
0.0	66.00	1P.00	72.00	0	0	0	-179.96	60.11
0.0	11A.00	-1P.00	72.00	0	0	0	-179.91	60.11
0.0	11A.00	1P.00	72.00	0	0	0	179.91	60.11
0.0	66.00	1P.00	72.00	0	0	0	179.91	60.22
45.00	66.00	-1P.00	72.00	0	0	0	179.76	60.10
45.00	11A.00	-1P.00	72.00	0	0	0	179.96	60.11
45.00	11A.00	1P.00	72.00	0	0	0	179.83	60.06

45.00	1P.00	72.00	0	0	0	60.14
90.07	56.70	-18.00	72.00	0	0	179.66
90.00	118.80	-1P.00	72.00	0	0	60.09
90.00	118.80	1P.00	72.00	0	0	179.85
90.00	118.80	1P.00	72.00	0	0	179.85
90.00	66.00	1P.00	72.00	0	0	60.01
135.60	66.00	-1P.00	72.00	0	0	179.60
135.00	118.80	-1P.00	72.00	0	0	59.95
135.00	118.80	1P.00	72.00	0	0	179.66
135.00	118.80	1P.00	72.00	0	0	179.65
135.00	118.80	1P.00	72.00	0	0	179.65
135.00	118.80	1P.00	72.00	0	0	179.66
180.20	66.00	-18.00	72.00	0	0	59.97
180.20	118.80	-18.00	72.00	0	0	179.78
180.20	118.80	1P.00	72.00	0	0	179.81
180.20	118.80	1P.00	72.00	0	0	179.81
180.20	118.80	1P.00	72.00	0	0	179.81
180.20	66.00	1P.00	72.00	0	0	59.97

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 3

SUN ELEVATION 60.00
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

IDENTIFICATION: LEFT FORWARD PANEL

XROT 150.00
 YROT 0.0
 ZROT 36.00
 DISTG 1000.00

A/C YAW

BOUNDARY POINTS -- BODY AXES
(STATIONLINE,BUTTLINE,WAERLINF)

	X	Y	Z	STATUS FLAGS FOR POINTS	INRFL	ISHAD	INTRF	GLINT	AZIMUTH	ELEVATION
-180.00	27.60	-18.00	42.00	1	0	0	0	-178.10	-59.74	
-180.00	54.00	-18.00	72.00	1	0	0	0	-178.09	-59.76	
-180.00	54.00	-18.00	42.00	1	0	0	0	-178.09	-59.79	
-135.00	27.60	-18.00	42.00	1	0	0	0	-90.47	-59.84	
-135.00	54.00	-18.00	72.00	1	0	0	0	-90.38	-59.84	
-135.00	54.00	-18.00	42.00	1	0	0	0	-90.38	-59.88	
-90.00	27.60	-18.00	42.00	1	0	0	0	1.42	-60.03	
-90.00	54.00	-18.00	72.00	1	0	0	0	1.54	-60.00	
-90.00	54.00	-18.00	42.00	1	1	0	0	1.54	-60.03	
-45.00	27.60	-18.00	42.00	1	0	0	0	89.65	-60.20	
-45.00	54.00	-18.00	72.00	1	0	0	0	89.74	-60.12	
-45.00	54.00	-18.00	42.00	1	0	0	0	89.74	-60.16	
0.0	27.60	-18.00	42.00	0	0	0	0	-179.91	-60.24	
0.0	54.00	-18.00	72.00	0	0	0	0	-179.91	-60.15	
0.0	54.00	-18.00	42.00	0	0	0	0	-179.91	-60.19	
45.00	27.60	-18.00	42.00	0	0	0	0	-89.53	-60.14	
45.00	54.00	-18.00	72.00	0	0	0	0	-89.62	-60.07	
45.00	54.00	-18.00	42.00	0	0	0	0	-89.62	-60.11	
90.00	27.60	-18.00	42.00	0	0	0	0	-1.41	-59.96	
90.00	54.00	-18.00	72.00	0	0	0	0	-1.54	-59.92	
90.00	54.00	-18.00	42.00	0	0	0	0	-1.54	-59.96	
135.00	27.60	-18.00	42.00	0	0	0	0	90.35	-59.79	
135.00	54.00	-18.00	72.00	0	0	0	0	91.26	-59.79	

135.00	54.00	-18.00	42.00	0	0	0	90.26	-59.62
180.00	27.60	-18.00	42.00	0	0	0	177.92	-59.74
160.00	54.00	-18.00	72.00	0	0	0	177.92	-59.76
189.00	54.00	-18.00	42.00	0	0	0	177.92	-59.79

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 4

IDENTIFICATION: RIGHT FORWARD PANEL

SUN ELEVATION	60.00
A/C PITCH ATTITUDE	0.0
A/C ROLL ATTITUDE	0.0

XRCT	150.00
YRCT	0.0
ZRCT	36.00
RTSTG	1000.00

A/C YAW

BOUNDARY POINTS -- BODY AXES	
(STATIONLINE-BUTTLINE-WATERLINE)	
X	Y
7	7

			STATUS FLAGS FOR POINTS		
			TNRL	ISHAD	INTPF
-180.00	27.60	18.00	92.00	0	0
-180.00	54.00	18.00	72.00	0	0
-180.00	54.00	18.00	42.00	0	0
-135.00	27.60	18.00	42.00	0	0
-135.00	54.00	18.00	72.00	0	0
-135.00	54.00	18.00	92.00	0	0
-90.00	27.60	18.00	42.00	0	0
-90.00	54.00	18.00	72.00	0	0
-90.00	54.00	18.00	92.00	0	0
-45.00	27.60	18.00	42.00	0	0
-45.00	54.00	18.00	72.00	0	0
-45.00	54.00	18.00	92.00	0	0
0.0	27.60	18.00	42.00	0	0
0.0	54.00	18.00	72.00	0	0
0.0	54.00	18.00	92.00	0	0
45.00	27.60	18.00	42.00	1	0
45.00	54.00	18.00	72.00	1	0
45.00	54.00	18.00	92.00	1	0
90.00	27.60	18.00	42.00	1	0
90.00	54.00	18.00	72.00	1	0
90.00	54.00	18.00	92.00	1	0
135.00	27.60	18.00	42.00	1	0
135.00	54.00	18.00	72.00	1	0

GLINT	
AZIMUTH	ELEVATION

135.00	54.00	19.00	42.00	1	0	0	90.38	-59.88
180.00	27.60	18.00	42.00	1	0	0	178.10	-59.74
180.00	54.00	19.00	72.00	1	0	0	178.09	-59.75
185.00	54.00	19.00	42.00	1	0	0	178.09	-59.79

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

*** SIM GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 4 IDENTIFICATION: LEFT AFT PANEL

SUN ELEVATION 60.00

A/C PITCH ATTITUDE 0.0

A/C ROLL ATTITUDE 0.0

XROT 150.00
YROT 0.0
ZROT 36.00
DISTG 1000.00A/C YAW
BOUNDARY POINTS - BODY AXES
(STATIONLINE BUTTLINE, WATERLINE)GLINT
AZIMUTH ELEVATION

A/C YAW	STATIONLINE	BUTTLINE	WATERLINE	IRFL	ISMAC	INTR	STATUS FLAGS FOR POINTS
-180.00	66.00	-18.00	72.00	1	0	0	-178.09 -59.76
-180.00	118.00	-18.00	72.00	1	0	0	-178.09 -50.69
-160.00	118.00	-18.00	42.00	1	0	0	-178.09 -59.93
-160.00	66.00	-18.00	42.00	1	0	0	-178.09 -59.82
-160.00	66.00	-18.00	72.00	1	0	0	-90.34 -59.86
-135.00	118.00	-18.00	72.00	1	0	0	-90.17 -59.94
-135.00	118.00	-18.00	42.00	1	0	0	-90.17 -59.97
-135.00	66.00	-18.00	42.00	1	0	0	-90.34 -59.90
-90.00	66.00	-18.00	72.00	1	0	0	1.60 -60.00
-90.00	118.00	-18.00	72.00	1	0	0	1.85 -60.00
-90.00	118.00	-18.00	42.00	1	0	0	1.85 -60.03
-90.00	66.00	-18.00	42.00	1	0	0	1.60 -60.03
-45.00	66.00	-18.00	42.00	1	0	0	89.70 -60.11
-45.00	118.00	-18.00	72.00	1	0	0	89.96 -60.06
-45.00	118.00	-18.00	42.00	1	0	0	89.70 -60.14
0.0	66.00	-18.00	72.00	0	0	0	-179.91 -60.13
0.0	118.00	-18.00	72.00	0	0	0	-179.91 -60.02
0.0	118.00	-18.00	42.00	0	0	0	-179.91 -60.06
0.0	66.00	-18.00	42.00	0	0	0	-179.91 -60.17
45.00	66.00	-18.00	72.00	0	0	0	-89.66 -60.05
45.00	118.00	-18.00	72.00	0	0	0	-89.83 -60.08
45.00	118.00	-18.00	42.00	0	0	0	-89.83 -60.01

45.00	66.00	-18.00	42.00	0	0	0	-89.66	-50.09
90.00	66.00	-18.00	72.00	0	0	0	-1.60	+79.52
90.00	118.00	-18.00	72.00	0	0	0	-1.85	-50.92
90.00	118.00	-18.00	72.00	0	0	0	-1.85	-50.92
90.00	118.00	-18.00	42.00	0	0	0	-1.85	-50.92
90.00	66.00	-18.00	42.00	0	0	0	-1.60	-59.96
90.00	66.00	-18.00	72.00	0	0	0	-1.60	-59.96
135.00	66.00	-18.00	72.00	0	0	0	-1.60	-59.96
135.00	118.00	-18.00	72.00	0	0	0	-1.60	-59.96
135.00	118.00	-18.00	42.00	0	0	0	-1.60	-59.96
135.00	66.00	-18.00	42.00	0	0	0	-1.60	-59.96
186.00	66.00	-18.00	72.00	0	0	0	177.92	-59.78
186.00	118.00	-18.00	72.00	0	0	0	177.92	-59.89
186.00	118.00	-18.00	42.00	0	0	0	177.92	-59.93
186.00	66.00	-18.00	42.00	0	0	0	177.92	-59.93

FLAT SURFACES WITH FENCES SAMPLE CASE ONE

*** SUM GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 6

IDENTIFICATION: RIGHT AFT FANFL

SUN ELEVATION 60.00
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

SUN ELEVATION 150.00
 YACR 0.0
 ZACR 36.00
 NISTC 1700.00

A/C YAW

BOUNDARY POINTS -- ECDW ANES
(STATIONLINE,BUTLINE,VALPLINR)
VSTATUS FLAGS FOR POINTS
INPL ISMAD INTPL

GLINT AZIMUTH ELEVATION

-180.00	66.00	10.00	72.00	1	9	0
-160.00	110.00	10.00	72.00	0	9	0
-140.00	110.00	10.00	92.00	0	0	0
-120.00	110.00	10.00	42.00	0	0	0
-100.00	66.00	10.00	72.00	0	0	0
-80.00	66.00	10.00	72.00	0	0	0
-60.00	110.00	10.00	72.00	0	0	0
-40.00	110.00	10.00	42.00	0	0	0
-20.00	66.00	10.00	72.00	0	0	0
0.00	66.00	10.00	72.00	0	0	0
20.00	110.00	10.00	42.00	0	0	0
40.00	66.00	10.00	72.00	0	0	0
60.00	110.00	10.00	72.00	0	0	0
80.00	66.00	10.00	72.00	0	0	0
100.00	110.00	10.00	72.00	0	0	0
120.00	110.00	10.00	92.00	0	0	0
140.00	110.00	10.00	42.00	0	0	0
160.00	66.00	10.00	72.00	0	0	0
180.00	66.00	10.00	72.00	0	0	0
200.00	110.00	10.00	42.00	0	0	0
220.00	66.00	10.00	72.00	0	0	0
240.00	110.00	10.00	72.00	0	0	0
260.00	66.00	10.00	72.00	0	0	0
280.00	110.00	10.00	42.00	0	0	0
300.00	66.00	10.00	72.00	0	0	0
320.00	110.00	10.00	72.00	0	0	0
340.00	66.00	10.00	42.00	0	0	0
360.00	110.00	10.00	72.00	0	0	0
380.00	66.00	10.00	72.00	0	0	0
400.00	110.00	10.00	42.00	0	0	0
420.00	66.00	10.00	72.00	0	0	0
440.00	110.00	10.00	72.00	0	0	0
460.00	66.00	10.00	42.00	0	0	0
480.00	110.00	10.00	72.00	0	0	0
500.00	66.00	10.00	72.00	0	0	0
520.00	110.00	10.00	42.00	0	0	0
540.00	66.00	10.00	72.00	0	0	0
560.00	110.00	10.00	72.00	0	0	0
580.00	66.00	10.00	42.00	0	0	0
600.00	110.00	10.00	72.00	0	0	0
620.00	66.00	10.00	72.00	0	0	0
640.00	110.00	10.00	42.00	0	0	0
660.00	66.00	10.00	72.00	0	0	0
680.00	110.00	10.00	72.00	0	0	0
700.00	66.00	10.00	42.00	0	0	0
720.00	110.00	10.00	72.00	0	0	0
740.00	66.00	10.00	72.00	0	0	0
760.00	110.00	10.00	42.00	0	0	0
780.00	66.00	10.00	72.00	0	0	0
800.00	110.00	10.00	72.00	0	0	0
820.00	66.00	10.00	42.00	0	0	0
840.00	110.00	10.00	72.00	0	0	0
860.00	66.00	10.00	72.00	0	0	0
880.00	110.00	10.00	42.00	0	0	0
900.00	66.00	10.00	72.00	0	0	0
920.00	110.00	10.00	72.00	0	0	0
940.00	66.00	10.00	42.00	0	0	0
960.00	110.00	10.00	72.00	0	0	0
980.00	66.00	10.00	72.00	0	0	0
1000.00	110.00	10.00	42.00	0	0	0

45.00	1P.00	42.00	1	0	0	-59.78	-60.14
50.00	56.70	1P.00	72.00	1	0	0	-1.60
50.00	11A.80	1P.00	72.00	1	0	0	-60.00
50.00	11A.80	1P.00	42.00	1	0	0	-1.85
50.00	66.00	1P.00	42.00	1	0	0	-60.00
50.00	66.00	1P.00	72.00	1	0	0	-1.85
50.00	11A.60	1P.00	72.00	1	0	0	-60.03
50.00	11A.60	1P.00	42.00	1	0	0	-1.60
50.00	66.00	1P.00	72.00	1	0	0	-60.03
50.00	11A.60	1P.00	72.00	1	0	0	-59.86
50.00	11A.60	1P.00	42.00	1	0	0	-59.86
50.00	11A.60	1P.00	42.00	1	0	0	-59.94
50.00	66.00	1P.00	42.00	1	0	0	-59.97
50.00	66.00	1P.00	72.00	1	0	0	-59.97
50.00	66.00	1P.00	72.00	1	0	0	-59.97
50.00	11A.80	1P.00	72.00	1	0	0	-59.97
50.00	11A.80	1P.00	42.00	1	0	0	-59.97
50.00	66.00	1P.00	42.00	1	0	0	-59.97
50.00	11A.80	1P.00	42.00	1	0	0	-59.97

FLAT SURFACES WITH FENCES

SAMPLE CASE ONE

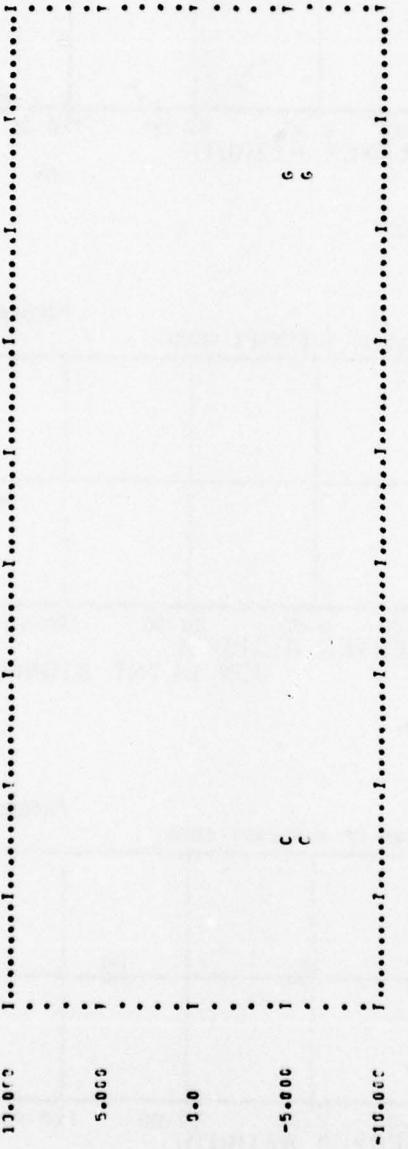
SUN ELEVATION: 60.00
 A/C PITCH ATTITUDE 7.0
 A/C ROLL ATTITUDE 0.0
 OBSERVER ANGLE DEGREES

OBSERVER ANGLE
DEGREES

17.000 16.000 15.000 14.000 13.000 12.000 11.000 10.000 9.000 8.000 7.000 6.000 5.000 4.000 3.000 2.000 1.000 0.000 -1.000 -2.000 -3.000 -4.000 -5.000 -6.000 -7.000 -8.000 -9.000 -10.000 -11.000 -12.000 -13.000 -14.000 -15.000 -16.000 -17.000

10.000 9.000 8.000 7.000 6.000 5.000 4.000 3.000 2.000 1.000 0.000 -1.000 -2.000 -3.000 -4.000 -5.000 -6.000 -7.000 -8.000 -9.000 -10.000 -11.000 -12.000 -13.000 -14.000 -15.000 -16.000 -17.000

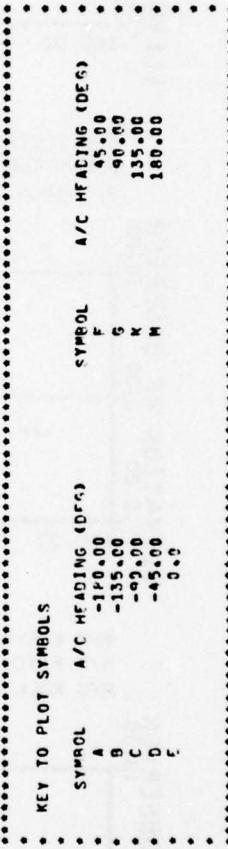
SUN GLINT SIGNATURE



-180.000 -160.000 -140.000 -120.000 -100.000 -80.000 -60.000 -40.000 -20.000 20.000 40.000 60.000 80.000 100.000 120.000 140.000 160.000 180.000

GLINT AZIMUTH WITH RESPECT TO SUN - DEGREES

PROBABILITY: 0.010

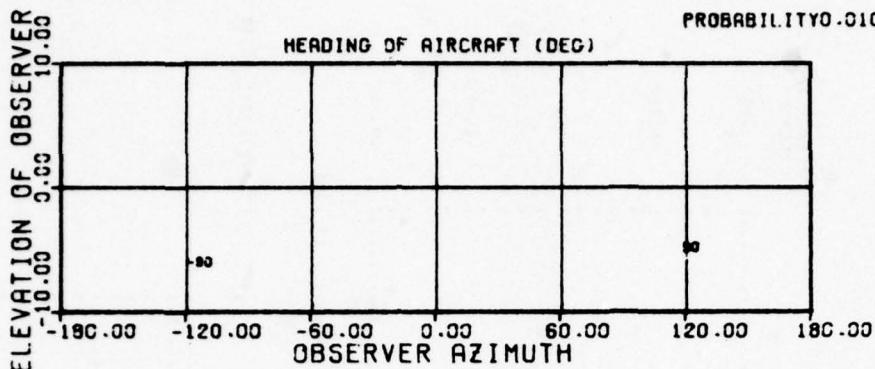


KEY TO PLOT SYMBOLS

SYMBOL	A/C HEADING (DEG)	SYMBOL	A/C HEADING (DEG)
A	-180.00	F	45.00
B	-135.00	G	90.00
C	-90.00	K	135.00
D	-45.00	M	180.00
E	0.00		

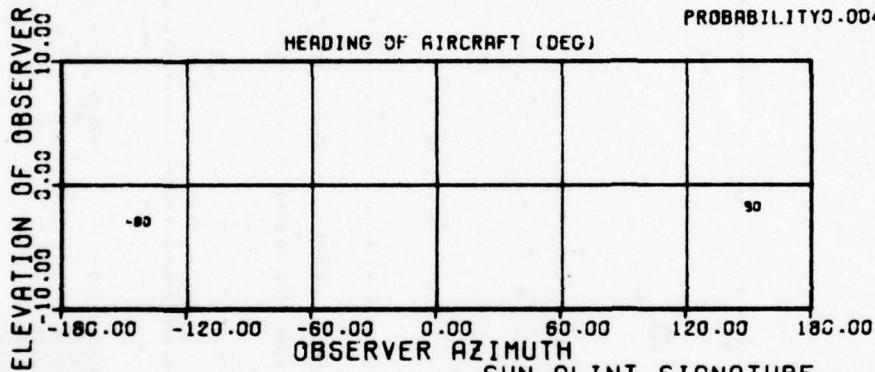
SUN ELEVATION ANGLE = 60.
A/C PITCH ATT = 0.0
A/C ROLL ATT = 0.0

PROBABILITY 0.010



SUN ELEVATION ANGLE = 30.
A/C PITCH ATT = 0.0
A/C ROLL ATT = 0.0

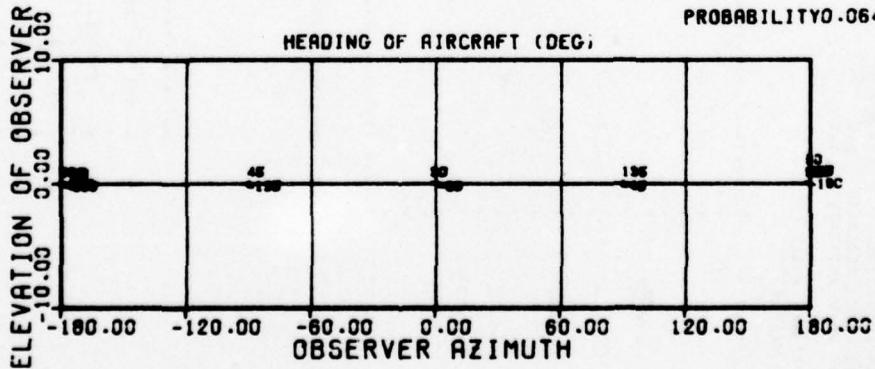
PROBABILITY 0.004



SUN GLINT SIGNATURE

SUN ELEVATION ANGLE = 0.
A/C PITCH ATT = 0.0
A/C ROLL ATT = 0.0

PROBABILITY 0.064



The second case represents a configuration with curved surfaces, shown in Figure A-3. The tabulation of points is shown in Table A-2. Some of the output data has been suppressed by setting PLOT = 2., no CALCOMP plots. Also, only one sun elevation, zero degrees, was chosen by setting GAMN = 1. and GAMI = 0. And since the glints from the windshield alone will vary from -180 to 180 degrees in azimuth at an aircraft heading of zero, only two headings were chosen to avoid much overlapping. The headings chosen were -45 and 45 degrees, by setting PSIN = 2., PSII = -45., and DPSI = 90. The card images for this case are shown in Figure A-4. The default option was not chosen for this case, in order to demonstrate the additional inputs required. These additional inputs, which specify which terms to use in the curve-fit analysis, come after the symmetry card for each reflective surface input except those that are symmetrical to the preceding surface. Additional engineering judgment must be exercised when using this option. For instance, the first and second surfaces appear to be a function of x , z , x^2 , y^2 , and z^2 . Therefore, a value of one is placed on the data cards corresponding to these inputs. The side panels seem to be independent of x and x^2 . Therefore, a value of one is placed on the data cards corresponding to z , y^2 , and z^2 locations only. The results from the inputs of Figure A-4 are shown on pages 123 through 138. It was found using these terms that the error for the curve-fit of the windshield was substantially higher than that for the other surfaces. This brings up a point. By inspecting the results of the curve-fit analysis, the program user can decide if one or more of the surfaces needs to be broken up into smaller pieces. A closer look at Figure A-3 reveals that part of the windshield actually looks like a side panel, thus independent of x and x^2 . If an imaginary line is drawn on the windshield connecting points $\textcircled{K}\textcircled{K}$ $\textcircled{D}\textcircled{D}$ \textcircled{D} , two surfaces are formed. Surface number one, which will be identified as WINDSHIELD FORWARD SECTION LEFT SIDE, will be formed from points \textcircled{D} \textcircled{E} \textcircled{F} \textcircled{G} \textcircled{H} \textcircled{K} $\textcircled{K}\textcircled{K}$ $\textcircled{D}\textcircled{D}$. The second surface, WINDSHIELD AFT SECTION LEFT SIDE, will be formed from points \textcircled{A} \textcircled{B} \textcircled{C} \textcircled{D} $\textcircled{D}\textcircled{D}$ $\textcircled{K}\textcircled{K}$. The modified inputs for case II with this decomposition of the windshield are shown in Figure A-5. The curve-fit result for the windshield as shown on page 145 is greatly improved; the complete results are on pages 139 through 158.

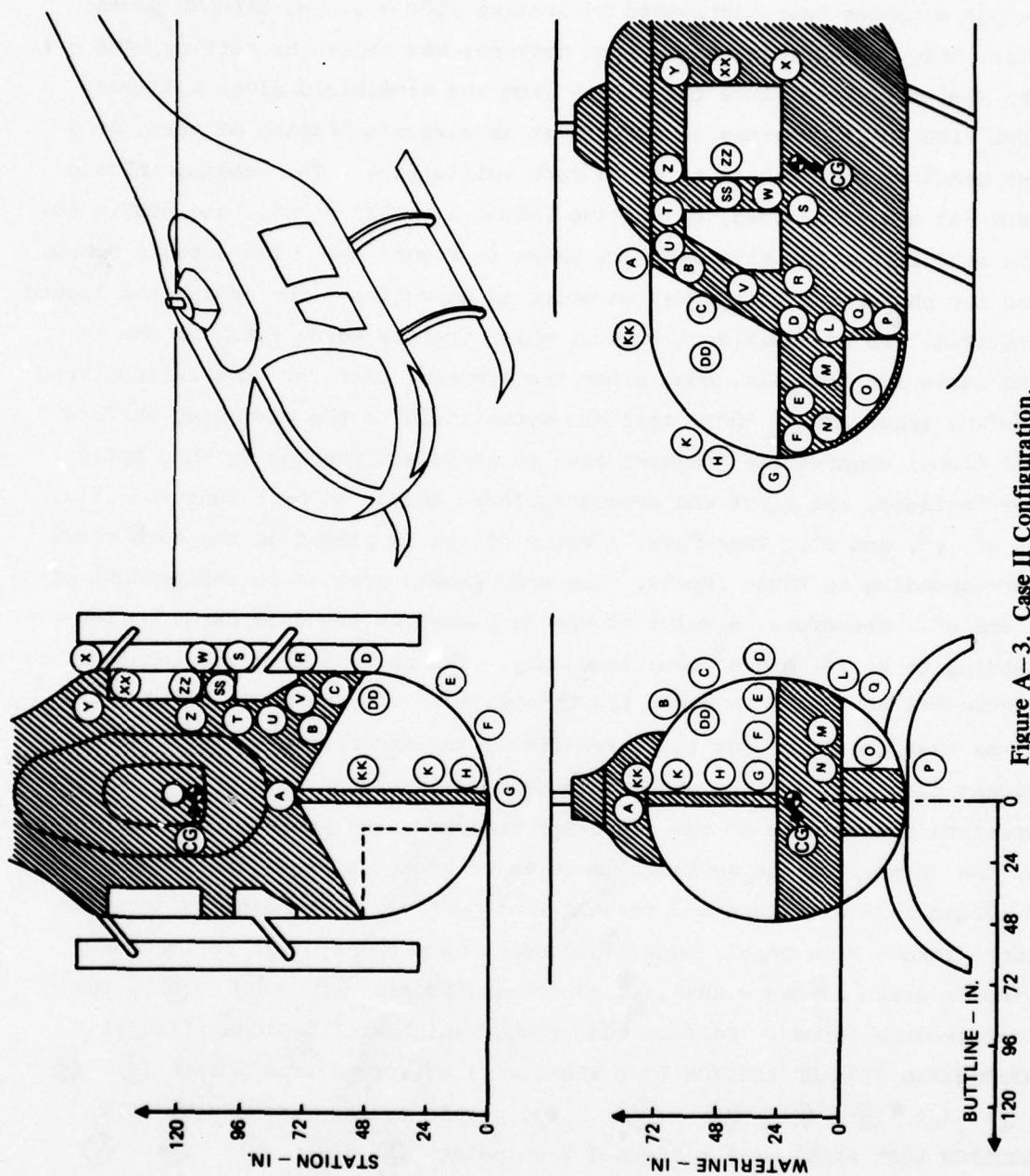


Figure A-3. Case II Configuration.

TABLE A-2. COORDINATES OF CASE II HELICOPTER CONFIGURATION

Coordinates of Center of Rotation (CG)

Station	Buttline	Waterline
114.	0	18.

Coordinates of Windshield Boundary Points

Station	Buttline	Waterline
(A)	72.	72.
(B)	66.	60.
(C)	60.	48.
(D)	48.	24.
(E)	16.25	24.
(F)	6.43	24.
(G)	0	24.
(H)	6.43	48.
(K)	16.25	60.
(KK)	48.	72.
(DD)	48.	54.

Coordinates of Lower Panel (Left Side) Boundary Points

Station	Buttline	Waterline
(L)	48.	0
(M)	30.	0
(N)	8.2	0
(O)	18.6	-12.
(P)	48.	-22.48
(Q)	48.	- 6.

Coordinates of Forward Side Panel (Left Side) Boundary Points

Station	Buttline	Waterline
(R)	72	24.
(S)	96.	24.
(SS)	96.	42.
(T)	96.	60.
(U)	84.	60.
(V)	72.	36.

TABLE A-2 - Continued

Coordinates of Aft Side Panel (Left Side) Boundary Points

Station	Buttline	Waterline
(W)	108.	-48.
(X)	144.	-48.
(XX)	144.	-44.5
(Y)	144.	-31.75
(Z)	108.	-31.75
(ZZ)	108.	-44.5

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

0.	2.	0.	0.		
1.	0.	0.	2.	-45.	90.
114.	0.	18.			
1500.					

0.	8.	2.	1.		
LEFT SIDE OF WINDSHIELD					

0.					
1.	0.	1.	0.	0.	0.
1.	1.	1.			
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.		
11.					
72.	0.	72.	66.	-31.75	60.
60.	-41.57	48.	48.	-48.	24.
16.25	-36.	24.	6.43	-24.	24.
0.	0.	24.	6.43	0.	48.
16.25	0.	60.	48.	0.	72.
48.	-37.47	54.			

RIGHT SIDE OF WINSHIELD

1.					
LOWER FRONT PANEL LEFT SIDE					

0.					
1.	0.	1.	0.	0.	0.
1.	1.	1.			
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.		
6.					
48.	-41.57	0.	30.	-37.47	0.
8.2	-12.	0.	18.6	-12.	-12.
48.	-12.	-22.48	48.	-37.47	-6.

LOWER FRONT PANEL RIGHT SIDE

1.					
FORWARD SIDE PANEL LEFT SIDE					

0.					
0.	0.	1.	0.	0.	0.
0.	1.	1.			
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.		
6.					
72.	-48.	24.	96.	-48.	24.
96.	-44.5	42.	96.	-31.75	60.
84.	-31.75	60.	72.	-46.48	36.

FORWARD SIDE PANEL RIGHT SIDE

1.					
AFT SIDE PANEL LEFT SIDE					

0.					
0.	0.	1.	0.	0.	0.
0.	1.	1.			
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.		
6.					
108.	-48.	24.	144.	-48.	24.
144.	-44.5	42.	144.	-31.75	60.
108.	-31.75	60.	108.	-44.5	42.

AFT SIDE PANEL RIGHT SIDE

1.

Figure A-4. Input for Case II; Card Images.

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO
 - MODIFIED WINDSHIELD

0.	2.	0.	0.		
1.	0.	0.	2.	-45.	90.
114.	0.	18.			
1500.	-10.	10.	-180.	180.	
0.	10.	2.	1.		

WINDSHIELD FORWARD SECTION LEFT SIDE

0.					
1.	0.	1.	0.	0.	0.
1.	1.	1.			
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.		
8.					
48.	-48.	24.	16.25	-36.	24.
6.43	-24.	24.	0.	0.	24.
6.43	0.	48.	16.25	0.	60.
48.	0.	72.	48.	-37.47	54.

WINDSHIELD FORWARD SECTION RIGHT SIDE

1.					
1.	WINDSHIELD AFT SECTION LEFT SIDE				
0.					
0.	0.	1.	0.	0.	0.
0.	1.	1.			
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.		
6.					
72.	0.	72.	66.	-31.75	60.
60.	-41.57	48.	48.	-48.	24.
48.	-37.47	54.	48.	0.	72.

WINDSHIELD AFT SECTION RIGHT SIDE

1.					
1.	LOWER FRONT PANEL LEFT SIDE				
0.					
1.	0.	1.	0.	0.	0.
1.	1.	1.			
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.		
6.					
48.	-41.57	0.	30.	-37.47	0.
8.2	-12.	0.	18.6	-12.	-12.
48.	-12.	-22.48	48.	-37.47	-6.

LOWER FRONT PANEL RIGHT SIDE

1.					
----	--	--	--	--	--

Figure A-5. Input for Case II (Windshield Modified); Card Images (Sheet 1 of 2).

FORWARD SIDE PANEL LEFT SIDE

0.	0.	1.	0.	0.	0.
0.	1.	1.	0.	0.	0.
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.
6.					
72.	-48.	24.	96.	-48.	24.
96.	-44.5	42.	96.	-31.75	60.
84.	-31.75	60.	72.	-46.48	36.

FORWARD SIDE PANEL RIGHT SIDE

1.					
AFT SIDE PANEL LEFT SIDE					
0.					
0.	0.	1.	0.	0.	0.
0.	1.	1.	0.	0.	0.
0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.
6.					
108.	-48.	24.	144.	-48.	24.
144.	-44.5	42.	144.	-31.75	60.
108.	-31.75	60.	108.	-44.5	42.

AFT SIDE PANEL RIGHT SIDE

1.

Figure A-5. Input for Case II (Windshield Modified); Card Images (Sheet 2 of 2).

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** INPUT DATA ***

```
**CONTROL OPTIONS
  DEFAULT OPTION      0.          PRINT OPTION      2.
  PLOT OPTION        2.          PROBABILITY OPT    1.

**AIRCRAFT INITIAL EULER ORIENTATION
  PITCH             0.0          ROLL            0.0          YAW            -45.0

**SELECTED YAW ANGLE ROTATIONS
  NO. OF ANGLES     2.          INITIAL ANGLE   -45.0          ANGLE INCREMENT  90.0

**SELECTED SUN ELEVATIONS
  NO. OF ANGLES     1.          INITIAL ANGLE   0.0          ANGLE INCREMENT  0.0

**REFERENCE INFORMATION FOR MEASURING SUN GLINT ANGLES
  REFERENCE POINT FOR MEASURING ANGLES OF GLINT SIGNATURE
    X                114.00        Y                2.          Z                16.00          DISTANCE AT WHICH GLINT SIGNATURE FORMED (FT.) 1500.000

**PLOT SCALING
  GLINT AZIMUTH
    MINIMUM          -180.0        GLINT ELEVATION
    MAXIMUM          180.0          MINIMUM         -10.0
                                         MAXIMUM         10.0

**BOUNDARIES FOR CALCULATING PROBABILITY
  GLINT AZIMUTH
    MINIMUM          -180.0        GLINT ELEVATION
    MAXIMUM          180.0          MINIMUM         -10.0
                                         MAXIMUM         10.0
```

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** INPUT DATA ***

*REFLECTIVE SURFACE DATA
NO. OF SURFACES 8.

*REFLECTIVE SURFACE NO. 1 IDENTIFICATION: LEFT SIDE OF WINDSHIELD

CURVE-FIT INPUT DATA
DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 1.
(2) Y TERM 0.
(3) Z TERM 1.
(4) XY TERM 1.
(5) XZ TERM 1.
(6) YZ TERM 1.
(7) XSQ TERM 1.
(8) YSQ TERM 1.
(9) ZSQ TERM 1.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0
(2) Y COEF 0.0
(3) Z COEF 0.0
(4) XY COEF 0.0
(5) XZ COEF 0.0
(6) YZ COEF 0.0
(7) XSQ COEF 0.0
(8) YSQ COEF 0.0
(9) ZSQ COEF 0.0
(10) CONSTANT 0.0

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

	X	Y	Z	X	Y	Z
1	72.00	0.0	72.00	{ 2 }	66.00	-31.75
2	48.00	-48.00	24.00	{ 5 }	16.25	-36.00
3	0.0	0.0	24.00	{ 8 }	6.43	0.0
4	48.00	0.0	72.00	{11}	48.00	-37.47

*REFLECTIVE SURFACE NO. 2 IDENTIFICATION: RIGHT SIDE OF WINDSHIELD

CURVE-FIT INPUT DATA
DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 1.
(2) Y TERM 0.
(3) Z TERM 1.
(4) XY TERM 1.
(5) XZ TERM 1.
(6) YZ TERM 1.
(7) XSQ TERM 1.
(8) YSQ TERM 1.
(9) ZSQ TERM 1.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0
(2) Y COEF 0.0
(3) Z COEF 0.0
(4) XY COEF 0.0
(5) XZ COEF 0.0
(6) YZ COEF 0.0
(7) XSQ COEF 0.0
(8) YSQ COEF 0.0
(9) ZSQ COEF 0.0
(10) CONSTANT 0.0

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

	X	Y	Z	X	Y	Z
1	72.00	0.0	72.00	{ 2 }	66.00	31.75
2	48.00	48.00	24.00	{ 5 }	16.25	36.00
3	0.0	0.0	24.00	{ 8 }	6.43	0.0
4	48.00	0.0	72.00	{11}	48.00	37.47

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** INPUT DATA ***

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: LOWER FRONT PANEL LEFT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM	1.	(2) Y TERM	0.
(6) Y2 TERM	0.	(7) XSQ TERM	1.

(3) Z TERM	1.	(4) XY TERM	0.
(8) YSO TERM	1.	(9) ZSQ TERM	1.

(5) X2 TERM 0.

(6) Y2 TERM 1.

(7) ZSQ TERM 1.

(8) YSO TERM 1.

(9) XY TERM 1.

(10) Z TERM 1.

(11) X COEF 0.0

(12) Y COEF 0.0

(13) Z COEF 0.0

(14) XY COEF 0.0

(15) Y2 COEF 0.0

(16) XSQ COEF 0.0

(17) YSO COEF 0.0

(18) CONSTANT 0.0

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.0	(2) Y COEF	0.0	(3) Z COEF	0.0
(4) XY COEF	0.0	(5) X2 COEF	0.0	(6) Y2 COEF	0.0
(7) XSO COEF	0.0	(8) YSO COEF	0.0	(9) ZSQ COEF	0.0
(10) CONSTANT	0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z	X	Y	Z
(1)	-48.00	-41.57	(2)	50.00	-37.47
(4)	18.60	-12.00	(5)	48.00	-12.00

*REFLECTIVE SURFACE NO. 4 IDENTIFICATION: LOWER FRONT PANEL RIGHT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM	1.	(2) Y TERM	0.
(6) Y2 TERM	0.	(7) XSQ TERM	1.

(3) Z TERM	1.	(4) XY TERM	0.
(8) YSO TERM	1.	(9) ZSQ TERM	1.

(5) X2 TERM 0.

(6) Y2 TERM 1.

(7) ZSQ TERM 1.

(8) YSO TERM 1.

(9) XY TERM 1.

(10) Z TERM 1.

(11) X COEF 0.0

(12) Y COEF 0.0

(13) Z COEF 0.0

(14) XY COEF 0.0

(15) Y2 COEF 0.0

(16) XSQ COEF 0.0

(17) YSO COEF 0.0

(18) CONSTANT 0.0

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.0	(2) Y COEF	0.0	(3) Z COEF	0.0
(4) XY COEF	0.0	(5) X2 COEF	0.0	(6) Y2 COEF	0.0
(7) XSO COEF	0.0	(8) YSO COEF	0.0	(9) ZSQ COEF	0.0
(10) CONSTANT	0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z	X	Y	Z
(1)	-48.00	41.57	(2)	30.00	37.47
(4)	18.60	12.00	(5)	48.00	12.00

*REFLECTIVE SURFACE NO. 5 IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM	0.	(2) Y TERM	0.
(6) Y2 TERM	0.	(7) XSQ TERM	0.

(3) Z TERM	1.	(4) XY TERM	0.
(8) YSO TERM	1.	(9) ZSQ TERM	1.

(5) X2 TERM 0.

(6) Y2 TERM 1.

(7) ZSQ TERM 1.

(8) YSO TERM 1.

(9) XY TERM 1.

(10) Z TERM 1.

(11) X COEF 0.0

(12) Y COEF 0.0

(13) Z COEF 0.0

(14) XY COEF 0.0

(15) Y2 COEF 0.0

(16) XSQ COEF 0.0

(17) YSO COEF 0.0

(18) CONSTANT 0.0

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.0	(2) Y COEF	0.0	(3) Z COEF	0.0
(4) XY COEF	0.0	(5) X2 COEF	0.0	(6) Y2 COEF	0.0
(7) XSO COEF	0.0	(8) YSO COEF	0.0	(9) ZSQ COEF	0.0
(10) CONSTANT	0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z	X	Y	Z
(1)	-48.00	41.57	(2)	30.00	37.47
(4)	18.60	12.00	(5)	48.00	12.00

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** INPUT DATA ***

•REFLECTIVE SURFACE NO. 1 IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.3	(2) Y COEF	0.0	(3) Z COEF	0.0
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSO COEF	0.0	(8) YSO COEF	0.0	(9) ZSO COEF	0.0
(10) CONSTANT	0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE+BUITLINE+WATERLINE)

X	Y	Z	X	Y	Z
(1) 72.00	-48.00	24.00	(2) 96.00	-48.00	24.00
(4) 96.00	-31.75	60.00	(5) 84.00	-31.75	60.00

•REFLECTIVE SURFACE NO. 6 IDENTIFICATION: FORWARD SIDE PANEL RIGHT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 1.	(4) XY TERM 0.	(5) XZ TERM 0.
(6) YZ TERM 0.	(7) XSO TERM 0.	(8) YSO TERM 1.	(9) ZSO TERM 1.	

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0	(4) XY COEF 0.0	(5) XZ COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0	(9) ZSO COEF 0.0	
(7) XSO COEF 0.0	(8) YSO COEF 0.0			
(10) CONSTANT 0.0				

BOUNDARY POINTS -- BODY AXES (STATIONLINE+BUITLINE+WATERLINE)

X	Y	Z	X	Y	Z
(1) 72.00	48.00	24.00	(2) 96.00	48.00	24.00
(4) 96.00	31.75	60.00	(5) 84.00	31.75	60.00

•REFLECTIVE SURFACE NO. 7 IDENTIFICATION: AFT SIDE PANEL LEFT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 1.	(4) XY TERM 0.	(5) XZ TERM 0.
(6) YZ TERM 0.	(7) XSO TERM 0.	(8) YSO TERM 1.	(9) ZSO TERM 1.	

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0	(4) XY COEF 0.0	(5) XZ COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0	(9) ZSO COEF 0.0	
(7) XSO COEF 0.0	(8) YSO COEF 0.0			
(10) CONSTANT 0.0				

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** INPUT DATA ***

*REFLECTIVE-SURFACE NO. 7 IDENTIFICATION: AFT SIDE PANEL LEFT SIDE

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

	X	Y	Z	X	Y	Z
(1)	108.00	48.00	24.00	(2)	144.00	-48.00
(4)	144.00	31.75	60.00	(5)	108.00	-31.75

*REFLECTIVE SURFACE NO. 8 IDENTIFICATION: AFT SIDE PANEL RIGHT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1)	X TERM	0.	(2)	Y TERM	0.	(3)	Z TERM	1.	(4)	XY TERM	0.	(5)	XZ TERM	0.
(6)	YZ TERM	0.	(7)	XSG TERM	0.	(8)	YSQ TERM	1.	(9)	ZSG TERM	1.	(10)	ZSQ TERM	1.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1)	X COEF	0.0	(2)	Y COEF	0.0	(3)	Z COEF	0.0
(4)	X1 COEF	0.0	(5)	X2 COEF	0.0	(6)	Y2 COEF	0.0
(7)	XSG COEF	0.0	(8)	YSQ COEF	0.0	(9)	ZSG COEF	0.0
(10)	CONSTANT	0.0						

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

	X	Y	Z	X	Y	Z
(1)	108.00	48.00	24.00	(2)	144.00	-48.00
(4)	144.00	31.75	60.00	(5)	108.00	-31.75

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** RESULTS OF CURVE-FIT ANALYSIS ***

*REFLECTIVE SURFACE DATA

*REFLECTIVE SURFACE NO. 1 IDENTIFICATION: LEFT SIDE OF WINDSHIELD

VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.4971E-01	(2) Y COEF	0.0	(3) Z COEF	-0.2472E-02
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSG COEF	0.2959E-03	(8) YSG COEF	0.1003E-03	(9) ZSG COEF	0.1596E-03
(10) CONSTANT	0.1000E 01				

ROOT MEAN SQUARE ERROR OF FIT= 0.6609E-01

*REFLECTIVE SURFACE NO. 2 IDENTIFICATION: RIGHT SIDE OF WINDSHIELD

VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.4971E-01	(2) Y COEF	0.0	(3) Z COEF	-0.2472E-02
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSG COEF	0.2959E-03	(8) YSG COEF	0.1003E-03	(9) ZSG COEF	0.1596E-03
(10) CONSTANT	0.1000E 01				

ROOT MEAN SQUARE ERROR OF FIT= 0.6609E-01

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: LOWER FRONT PANEL LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.6519E-01	(2) Y COEF	0.0	(3) Z COEF	-0.5735E-02
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSG COEF	0.4787E-03	(8) YSG COEF	0.4786E-03	(9) ZSG COEF	0.4786E-03
(10) CONSTANT	0.1000E 01				

ROOT MEAN SQUARE ERROR OF FIT= 0.2958E-04

*REFLECTIVE SURFACE NO. 4 IDENTIFICATION: LOWER FRONT PANEL RIGHT SIDE

VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF	0.6319E-01	(2) Y COEF	0.0	(3) Z COEF	-0.5735E-02
(4) XY COEF	0.0	(5) XZ COEF	0.0	(6) YZ COEF	0.0
(7) XSG COEF	0.7787E-03	(8) YSG COEF	0.4786E-03	(9) ZSG COEF	0.4786E-03
(10) CONSTANT	0.1000E 01				

ROOT MEAN SQUARE ERROR OF FIT= 0.2958E-04

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** RESULTS OF CURVE-FIT ANALYSIS ***

•REFLECTIVE SURFACE DATA

•REFLECTIVE SURFACE NO. 5 IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.0 (2) Y COEF 0.0 (3) Z COEF -0.5312E-02
 (4) XY COEF 0.0 (5) XZ COEF 2.0 (6) YZ COEF 0.0
 (7) XSG COEF 0.0 (8) YSG COEF 0.4410E-03 (9) ZSG COEF 0.4414E-03
 (10) CONSTANT -0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.2322E-04

•REFLECTIVE SURFACE NO. 6 IDENTIFICATION: FORWARD SIDE PANEL RIGHT SIDE

VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.0 (2) Y COEF 0.0 (3) Z COEF -0.5312E-02
 (4) XY COEF 0.0 (5) XZ COEF 0.0 (6) YZ COEF 0.0
 (7) XSG COEF 0.0 (8) YSG COEF 0.4410E-03 (9) ZSG COEF 0.4414E-03
 (10) CONSTANT -0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.2322E-04

•REFLECTIVE SURFACE NO. 7 IDENTIFICATION: AFT SIDE PANEL LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.0 (2) Y COEF 0.0 (3) Z COEF -0.5307E-02
 (4) XY COEF 0.0 (5) XZ COEF 0.5 (6) YZ COEF 0.0
 (7) XSG COEF 0.0 (8) YSG COEF 0.4410E-03 (9) ZSG COEF 0.4413E-03
 (10) CONSTANT -0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.2322E-04

•REFLECTIVE SURFACE NO. 8 IDENTIFICATION: AFT SIDE PANEL RIGHT SIDE

VALUES FOR COEFFICIENTS OF SURFACE
 (1) X COEF 0.0 (2) Y COEF 0.0 (3) Z COEF -0.5307E-02
 (4) XY COEF 0.0 (5) XZ COFF 0.0 (6) YZ COFF 0.0
 (7) XSG COEF 0.0 (8) YSG COEF 0.4410E-03 (9) ZSG COEF 0.4413E-03
 (10) CONSTANT -0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.4186E-15

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 1

IDENTIFICATION: LEFT SIDE OF WINDSHIELD

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.0
A/C ROLL ATTITUDE	0.0

XROT	114.00
YROT	0.0
ZROT	16.00
DISTG	1500.00

A/C YAW

BOUNDARY POINTS -- BODY AXES
(STATIONLINE,BUTTLINE,WATERLINE)

AZIMUTH GLINT ELEVATION

X	Y	Z	STATUS FLAGS FOR POINTS	IPRFL	ISHAO	INTRF		
-45.00	72.00	0.0	72.00	1	0	0	-127.88	-46.47
-45.00	66.00	-31.75	60.00	1	0	0	-9.67	-62.74
-45.00	60.00	-41.57	48.00	1	0	0	29.73	-29.76
-45.00	48.00	-48.00	24.00	1	0	0	71.22	2.86
-45.00	16.25	-36.00	24.00	0	0	0	-173.25	-0.16
-45.00	6.43	+24.00	24.00	0	0	0	-136.68	-1.01
-45.00	0.0	0.0	24.00	0	0	0	-89.77	-1.82
-45.00	6.43	0.0	48.00	0	0	0	-96.39	25.64
-45.00	16.25	0.0	60.00	0	0	0	-117.14	-32.18
-45.00	48.00	0.0	72.00	1	0	0	-177.52	-15.79
-45.00	48.00	-37.47	54.00	1	0	0	89.12	-48.95
-45.00	72.00	0.0	72.00	1	0	0	127.88	-46.47
-45.00	66.00	-31.75	60.00	1	0	0	175.62	-2.53
-45.00	60.00	-41.57	48.00	0	0	0	-157.47	8.69
-45.00	48.00	-48.00	24.00	0	0	0	-108.53	-2.01
-45.00	15.25	-36.00	24.00	0	0	0	6.55	-3.09
-45.00	6.43	-24.00	24.00	1	0	0	12.96	-2.58
-45.00	0.0	0.0	24.00	0	0	0	59.77	-1.82
-45.00	6.43	0.0	46.00	0	0	0	96.39	25.64
-45.00	16.25	0.0	60.00	0	0	0	117.14	42.15
-45.00	48.00	0.0	72.00	1	0	0	177.52	-15.79
-45.00	48.00	-37.47	54.00	1	J	0	-130.87	29.57

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 2 IDENTIFICATION: RIGHT SIDE OF WINSHIELD

*** SUM GLINT SIGNATURE ***

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.0
A/C ROLL ATTITUDE	0.0

XROT	114.00
YROT	-0.0
ZROT	18.00
DISTG	1500.00

A/C YAW	BOUNDARY POINTS -- BODY AXES (STATIONLINE,BULLTLINE,WATERLINE)			STATUS FLAGS FOR POINTS INRFL ISHAD INTRF			GLINT AZIMUTH ELEVATION
	X	Y	Z				
-45.00	72.00	0.0	72.00	1	0	0	-127.88 -94.47
-45.00	66.00	31.75	60.00	1	0	0	-175.62 -2.53
-45.00	60.00	41.57	48.00	0	0	0	157.47 8.89
-45.00	48.00	48.00	24.00	0	0	0	108.53 -2.01
-45.00	16.25	36.00	24.00	0	0	0	-6.55 -3.09
-45.00	6.43	24.00	24.00	0	0	0	-42.96 -2.58
-45.00	3.0	0.0	24.00	0	0	0	-89.77 -1.82
-45.00	6.43	0.0	48.00	0	0	0	-36.39 25.64
-45.00	16.25	0.0	60.00	0	0	0	-117.14 -92.15
-45.00	48.00	0.0	72.00	1	0	0	-177.52 -15.79
-45.00	48.00	37.47	54.00	0	0	0	130.87 29.57
-45.00	72.00	0.0	72.00	1	0	0	-127.88 -94.47
45.00	66.00	31.75	60.00	1	0	0	9.67 -62.74
45.00	60.00	41.57	48.00	1	0	0	-29.73 -40.74
45.00	48.00	48.00	24.00	1	0	0	-71.22 2.86
45.00	16.25	36.00	24.00	0	0	0	173.25 -0.16
45.00	6.43	24.00	24.00	0	0	0	136.68 -1.01
45.00	0.0	0.0	24.00	0	0	0	89.77 -1.82
45.00	6.43	0.0	48.00	0	0	0	96.39 25.64
45.00	16.25	0.0	60.00	0	0	0	117.14 92.15
45.00	48.00	0.0	72.00	1	0	0	177.52 -15.79
45.00	48.00	37.47	54.00	1	0	0	-9.12 -46.95

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 3

IDENTIFICATION: LOWER FRONT PANEL LEFT SIDE

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.6
A/C ROLL ATTITUDE	0.0

XROT	114.00
YROT	0.0
ZROT	16.00
DISTG	1500.00

A/C YAW	BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)			STATUS FLAGS FOR POINTS		
	X	Y	Z	INRFL	ISHAD	INTRF
-45.00	48.00	-41.57	0.0	1	0	0
-45.00	30.00	-37.47	0.0	1	0	0
-45.00	8.20	-12.00	0.0	0	0	0
-45.00	18.60	-12.00	-12.00	0	0	0
-45.00	48.00	-12.00	-22.48	1	0	0
-45.00	48.00	-37.47	-6.00	1	0	0
-45.00	48.00	-41.57	0.0	0	0	0
-45.00	30.00	-37.47	0.0	0	0	0
-45.00	8.20	-12.00	0.0	0	0	0
-45.00	18.60	-12.00	-12.00	0	0	0
-45.00	48.00	-12.00	-22.48	0	0	0
-45.00	48.00	-37.47	-6.00	0	0	0
-45.00	48.00	-41.57	0.0	0	0	0
-45.00	30.00	-37.47	0.0	0	0	0
-45.00	8.20	-12.00	0.0	0	0	0
-45.00	18.60	-12.00	-12.00	0	0	0
-45.00	48.00	-12.00	-22.48	0	0	0
-45.00	48.00	-37.47	-6.00	0	0	0

GLINT	AZIMUTH	ELEVATION
108.44	37.84	
150.08	16.68	
-136.41	-26.21	
-159.91	-22.72	
116.35	19.96	
122.71	43.70	
-108.22	-37.85	
-54.12	-54.77	
-75.03	-49.61	
129.00	-66.18	
-176.29	-20.19	
-122.49	-43.75	

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 4

IDENTIFICATION: LOWER FRONT PANEL RIGHT SIDE

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

A/C YAW

BOUNDARY POINTS -- BODY AXES
 (STATIONLINE,PUTTLINE,WATERLINE)
 X Y Z

				STATUS FLAGS FOR POINTS			
				INRFL	ISHAD	INTRF	GLINT
							AZIMUTH ELEVATION
-45.00	48.00	41.57	0.0	0	0	0	108.22 -37.85
-45.00	30.00	37.47	0.0	0	0	0	54.12 -54.77
-45.00	8.20	12.00	0.0	0	0	0	-75.03 -49.61
-45.00	18.60	12.00	-12.00	0	0	0	-129.00 -66.18
-45.00	48.00	12.00	-22.48	0	0	0	176.29 -20.19
-45.00	48.00	37.47	-6.00	0	0	0	122.49 -43.75
45.00	48.00	41.57	0.0	1	0	0	-108.44 37.84
45.00	30.00	37.47	0.0	1	0	0	-150.88 16.68
45.00	8.20	12.00	0.0	0	0	0	136.91 -24.21
45.00	18.60	12.00	-12.00	0	0	0	159.91 -22.72
45.00	48.00	12.00	-22.48	1	0	0	-176.35 19.96
45.00	48.00	37.47	-6.00	1	0	0	-122.71 -43.70

*** SUN GLINT SIGNATURE ***

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 5 IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

A/C YAW

BOUNDARY POINTS -- BODY AXES
 (STATIONLINE,BUTTLINE,WATERLINE)
 X Y Z

	X	Y	Z	STATUS FLAGS FOR POINTS INRFL ISHAD INTRF	GLINT AZIMUTH ELEVATION
-45.00	72.00	-48.00	24.00	1 0 0	90.01 0.05
-45.00	96.00	-48.00	24.03	1 0 0	90.07 0.05
-45.00	96.00	-49.50	42.00	1 0 0	99.37 -29.44
-45.00	96.00	-31.75	60.00	1 0 0	142.26 -44.50
-45.00	84.00	-31.75	60.00	1 0 0	142.26 -44.52
-45.00	72.00	-46.48	36.00	1 0 0	93.83 -20.02
-45.00	72.00	-48.00	24.00	0 0 0	-89.80 -0.01
-45.00	96.00	-48.00	24.00	0 0 0	-89.85 -0.01
-45.00	96.00	-44.50	42.00	0 0 0	-95.14 -29.49
-45.00	96.00	-31.75	60.00	0 0 0	-142.05 44.65
-45.00	84.00	-31.75	60.00	0 0 0	-142.06 44.67
-45.00	72.00	-46.48	36.00	0 0 0	-93.81 -20.06

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 6 IDENTIFICATION: FORWARD SIDE PANEL RIGHT SIDE

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

XROT 114.00
 YROT -0.0
 ZROT 16.00
 DISTG 1500.00

A/C YAW
 (STATIONLINE, BUTTLINE, WATERLINE)
 X Y Z
 (BODY AXES)

A/C YAW	BOUNDARY POINTS -- BODY AXES			STATUS FLAGS FOR POINTS			GLINT AZIMUTH ELEVATION
	INRFL	ISHAN	TNPF	INRFL	ISHAN	TNPF	
-45.00	72.00	48.00	24.00	0	0	0	89.80 -0.01
-45.00	96.00	48.00	24.00	0	0	0	39.85 -0.01
-45.00	96.00	44.50	42.00	0	0	0	99.14 29.49
-45.00	96.00	31.75	60.00	0	0	0	142.05 44.65
-45.00	84.00	31.75	60.00	0	0	0	142.06 44.67
-45.00	72.00	46.48	36.00	0	0	0	93.61 20.06
45.00	72.00	48.00	24.00	1	0	0	-90.01 0.05
45.00	96.00	48.00	24.00	1	0	0	-90.07 0.05
45.00	96.00	44.50	42.00	1	0	0	-99.37 -29.44
45.00	96.00	31.75	60.00	1	0	0	-142.26 -44.50
45.00	84.00	31.75	60.00	1	0	0	-142.26 -44.52
45.00	72.00	46.48	36.00	1	0	0	-93.83 -20.02

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 7

IDENTIFICATION: AFT SIDE PANEL LEFT SIDE

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.0
A/C ROLL ATTITUDE	0.0

XROT	114.00
YROT	0.0
ZROT	18.00
DISTG	1500.00

A/C YAW

(STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z	BOUNDARY POINTS -- BODY AXES			STATUS FLAGS FOR POINTS			AZIMUTH	GLINT ELEVATION
			TWRL	ISHAD	INTRF	TWRL	ISHAD	INTRF		
-45.00	108.00	-48.00	24.00	1	0	0	0	0	90.09	0.04
-45.00	144.00	-48.00	24.00	1	0	0	0	0	90.18	0.04
-45.00	144.00	-44.50	42.00	1	0	0	0	0	39.46	29.39
-45.00	144.00	-31.75	60.00	1	0	0	0	0	142.23	-44.43
-45.00	108.00	-31.75	60.00	1	0	0	0	0	142.24	-44.49
-45.00	108.00	-44.50	42.00	1	0	0	0	0	99.40	29.43
45.00	108.00	-48.00	24.00	0	0	0	0	0	-89.88	-0.08
45.00	144.00	-48.00	24.00	0	0	0	0	0	-89.96	-0.00
45.00	144.00	-44.50	42.00	0	0	0	0	0	-99.23	29.43
45.00	144.00	-31.75	60.00	0	0	0	0	0	-142.03	44.58
45.00	108.00	-31.75	60.00	0	0	0	0	0	-142.04	44.63
45.00	108.00	-44.50	42.00	0	0	0	0	0	-99.17	29.47

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 8

IDENTIFICATION: AFT SIDE PANEL RIGHT SIDE

*** SUN GLINT SIGNATURE ***

SUN ELEVATION

0.0

A/C PITCH ATTITUDE

0.0

A/C ROLL ATTITUDE

0.0

XROT	110.00
YROT	0.0
ZROT	18.00
DISTG	1500.00

A/C YAW	GLINT
	AZIMUTH ELEVATION

BOUNDARY POINTS -- BODY AXES	
(STATIONLINE,BUTTLINE,WATERLINE)	
X	Y
2	2

A/C YAW	STATUS FLAGS FOR POINTS			GLINT	AZIMUTH ELEVATION
	IMFL	ISHD	INTRF		
-45.00	108.00	48.00	24.00	0	0
-45.00	144.00	48.00	24.00	0	0
-45.00	144.00	44.50	42.00	0	0
-45.00	144.00	31.75	60.00	0	0
-45.00	108.00	31.75	60.00	0	0
-45.00	108.00	44.50	42.00	0	0
45.00	108.00	48.00	24.00	1	0
45.00	144.00	48.00	24.00	1	0
45.00	144.00	44.50	42.00	1	0
45.00	144.00	31.75	60.00	1	0
45.00	108.00	31.75	60.00	1	0
45.00	108.00	44.50	42.00	1	0

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** INPUT DATA ***

**CONTROL OPTIONS	PRINT OPTION	2.	PRINT OPTION OPT	2.	
DEFAULT OPTION	1.	PILOT OPTION	1.		
**AIRCRAFT INITIAL EULER ORIENTATION					
PITCH	0.0	ROLL	0.0	YAW	
NO. OF ANGLES	2.	INITIAL ANGLE	-45.0	ANGLE INCREMENT	-45.0
**SELECTED SUN ELEVATIONS					
NO. OF ANGLES	1.	INITIAL ANGLE	0.0	ANGLE INCREMENT	0.0
**REFERENCE POINT FOR MEASURING SUN GLINT ANGLES					
REFERENCE POINT FOR MEASURING ANGLES OF GLINT SIGNATURE				DISTANCE AT WHICH GLINT SIGNATURE FORMED (FT.)	1500.000
X	114.00	Y	16.00		
Z	0.0				
**PILOT SCALING					
GLINT AZIMUTH				CLINT ELEVATION	
MINIMUM	-180.0	MAXIMUM	180.0	MINIMUM	-10.0
MAXIMUM	180.0			MAXIMUM	10.0
**BOUNDARIES FOR CALCULATING PROBABILITY					
GLINT AZIMUTH				GLINT ELEVATION	
MINIMUM	-180.0	MAXIMUM	180.0	MINIMUM	-10.0
MAXIMUM	180.0			MAXIMUM	10.0

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** INPUT DATA ***

REFLECTIVE SURFACE DATA
NO. OF SURFACES 10.

REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD FORWARD SECTION LEFT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 1.	(2) Y TERM 0.
(6) Y2 TERM 0.	(7) XSG TERM 1.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0
(7) XSG COEF 0.0	(8) YSG COEF 0.0
(19) CONSTANT 0.0	

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

	X	Y	Z	X	Y	Z
1	48.00	-48.00	24.00	12.2	16.25	-36.00
4	0.0	0.0	24.00	(5)	6.43	0.0
7	48.00	0.0	72.00	(8)	48.00	-17.47

REFLECTIVE SURFACE NO. 2 IDENTIFICATION: WINDSHIELD FORWARD SECTION RIGHT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 1.	(2) Y TERM 0.
(6) Y2 TERM 0.	(7) XSG TERM 1.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0
(7) XSG COEF 0.0	(8) YSG COEF 0.0
(19) CONSTANT 0.0	

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

	X	Y	Z	X	Y	Z
1	48.00	48.00	24.00	12.2	16.25	36.00
4	0.0	0.0	24.00	(5)	6.43	0.0
7	48.00	0.0	72.00	(8)	48.00	37.47

REFLECTIVE SURFACE NO. 3 IDENTIFICATION: WINDSHIELD AFT SECTION LEFT SIDE

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** INPUT DATA ***

REFLECTIVE SURFACE NO. 3 IDENTIFICATION: WINDSHIELD AFT SECTION LEFT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 1.	(4) XY TERM 0.	(5) XZ TERM 1.
(6) YZ TERM 0.	(7) XSG TERM 0.	(8) YSG TERM 0.	(9) ZSG TERM 0.	

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0	(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0	(7) XSG COEF 0.0	(8) YSG COEF 0.0	(9) ZSG COEF 0.0
(10) CONSTANT 0.0								

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z	X	Y	Z	X	Y	Z
(1) 72.00	0.0	72.00	(2) 66.00	-31.75	60.00	(3) 60.00	-41.57	48.00
(4) 48.00	-48.00	24.00	(5) 48.00	-37.47	54.00	(6) 48.00	-37.47	72.00

REFLECTIVE SURFACE NO. 4 IDENTIFICATION: WINDSHIELD AFT SECTION RIGHT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 1.	(4) XY TERM 0.	(5) XZ TERM 1.
(6) YZ TERM 0.	(7) XSG TERM 0.	(8) YSG TERM 0.	(9) ZSG TERM 0.	

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0	(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0	(7) XSG COEF 0.0	(8) YSG COEF 0.0	(9) ZSG COEF 0.0
(10) CONSTANT 0.0								

BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)

X	Y	Z	X	Y	Z	X	Y	Z
(1) 72.00	0.0	72.00	(2) 66.00	31.75	60.00	(3) 60.00	41.57	48.00
(4) 48.00	48.00	24.00	(5) 48.00	37.47	54.00	(6) 48.00	37.47	72.00

REFLECTIVE SURFACE NO. 5 IDENTIFICATION: LOWER FRONT PANEL LEFT SIDE

CURVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 1.	(2) Y TERM 0.	(3) Z TERM 1.	(4) XY TERM 0.	(5) XZ TERM 1.
(6) YZ TERM 0.	(7) XSG TERM 0.	(8) YSG TERM 0.	(9) ZSG TERM 0.	

REFLECTIVE SURFACE NO. 6 IDENTIFICATION: LOWER FRONT PANEL RIGHT SIDE

BIG

CURVED PANELS WITH NO FORCES SAMPLE CASE TWO

*** INPUT DATA ***

•REFLECTIVE SURFACE NO. 5 IDENTIFICATION: LOWER FRONT PANEL LEFT SIDE

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0
(7) YSO COEF 0.0	(8) YSG COEF 0.0	(9) ZSO COEF 0.0
(10) CONSTANT 0.0		

BOUNDARY POINTS -- BODY AXES (STATION/LINE,PUTTLINE,WATERLINE)

X 2	Y 2	Z 2
(1) 48.00 -91.57 0.0	(2) 30.00 -27.47 0.0	
(4) 18.60 -12.00 -12.00	(5) 48.00 -12.00 -22.48	

•REFLECTIVE SURFACE NO. 6 IDENTIFICATION: LOWER FRONT FAIRFL RIGHT SIDE

CUEVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 1.	(2) Y TERM 0.	(3) Z TERM 1.
(4) YZ TERM 0.	(5) XSO TERM 1.	(6) YSG TERM 1.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 1.0	(2) Y COEF 0.0	(3) Z COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0
(7) YSO COEF 0.0	(8) YSG COEF 0.0	(9) ZSO COEF 0.0
(10) CONSTANT 0.0		

BOUNDARY POINTS -- BODY AXES (STATION/LINE,PUTTLINE,WATERLINE)

X 2	Y 2	Z 2
(1) 48.00 -91.57 0.0	(2) 30.00 -27.47 0.0	
(4) 18.60 12.00 -12.00	(5) 48.00 12.00 -22.48	

•REFLECTIVE SURFACE NO. 7 IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

CUEVE-FIT INPUT DATA

DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 1.
(4) YZ TERM 0.	(5) XSO TERM 0.	(6) YSG TERM 1.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0
(7) YSO COEF 0.0	(8) YSG COEF 0.0	(9) ZSO COEF 0.0
(10) CONSTANT 0.0		

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** INPUT DATA ***

*REFLECTIVE SURFACE NO. 7 IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

BOUNDARY POINTS -- BODY AXES (STATIONLINE-BUTTLINE-WATERLINE)

X	2	Z
Y		
(1) 72.00	-48.00	24.00
(4) 96.00	-31.75	60.00

*REFLECTIVE SURFACE NO. 8 IDENTIFICATION: FORWARD SIDE PANEL RIGHT SIDE

CURVE-FIT INPUT DATA

DESIRFC TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 0.
(6) X2 TERM 0.	(7) XSG TERM 0.	(8) YSG TERM 0.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0
(7) XSG COEF 0.0	(8) YSG COEF 0.0	(9) ZSG COEF 0.0

BOUNDARY POINTS -- BODY AXES (STATIONLINE-BUTTLINE-WATERLINE)

X	2	Z
Y		
(1) 72.00	48.00	24.00
(4) 96.00	-31.75	60.00

*REFLECTIVE SURFACE NO. 9 IDENTIFICATION: AFT SIDE PANEL LEFT SIDE

CURVE-FIT INPUT DATA

DESIRFC TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 0.
(6) X2 TERM 0.	(7) XSG TERM 0.	(8) YSG TERM 0.

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0
(7) XSG COEF 0.0	(8) YSG COEF 0.0	(9) ZSG COEF 0.0

BOUNDARY POINTS -- BODY AXES (STATIONLINE-BUTTLINE-WATERLINE)

X	2	Z
Y		
(1) 108.00	-48.00	24.00
(4) 144.00	-31.75	60.00

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*REFLECTIVE SURFACE NO. 10 IDENTIFICATION: AFT SIDE PANEL RIGHT SIDE
 CURVE-FIT INPUT DATA
 DESIRED TERMS FOR CURVE-FITTING

(1) X TERM 0.	(2) Y TERM 0.	(3) Z TERM 1.	(4) XY TERM 0.	(5) XZ TERM 1.	(6) YZ TERM 0.
(7) YZ TERM 0.	(8) YSG TERM 0.	(9) YSG TERM 1.	(10) ZSG TERM 0.		

INPUT VALUES FOR COEFFICIENTS OF SURFACE

(1) X COEF 0.0	(2) Y COEF 0.0	(3) Z COEF 0.0	
(4) XY COEF 0.0	(5) XZ COEF 0.0	(6) YZ COEF 0.0	
(7) YSG COEF 0.0	(8) YSG COEF 0.0	(9) ZSG COEF 0.0	
(10) CONSTANT 0.0			

BOUNDARY POINTS -- BODY AXES (STATION,LINE,RIGHT,LIN#,WATERLIN#)

X	Y	Z	X	Y	Z	X	Y	Z
(1) 100.00	40.00	40.00	(2) 144.00	48.00	24.00	(3) 144.00	44.50	22.00
(4) 144.00	31.75	60.00	(5) 108.00	31.75	60.00	(6) 108.00	44.50	42.00

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** RESULTS OF CURVE-FIT ANALYSIS ***

*REFLECTIVE SURFACE DATA

*REFLECTIVE SURFACE NO. 1 IDENTIFICATION: WINDSHIELD FORWARD SECTION LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.6320E-01
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XSO COEF	0.4788E-03
(12) CONSTANT	0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.6116E-05

*REFLECTIVE SURFACE NO. 2 IDENTIFICATION: WINDSHIELD FORWARD SECTION RIGHT SIDE

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.6320E-02
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XSO COEF	0.4788E-03
(12) CONSTANT	0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.6116E-05

*REFLECTIVE SURFACE NO. 3 IDENTIFICATION: WINDSHIELD AFT SECTION LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.0
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XSO COEF	0.0
(12) CONSTANT	-0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.9869E-05

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.0
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XSO COEF	0.0
(12) CONSTANT	-0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.9869E-05

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.0
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XSO COEF	0.0
(12) CONSTANT	0.4409E-03

ROOT MEAN SQUARE ERROR OF FIT= 0.9869E-05

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** RESULTS OF CURVE-FIT ANALYSIS ***

REFLECTIVE SURFACE DATA

*REFLECTIVE SURFACE NO. 5 IDENTIFICATION: LOWER FRONT PANEL LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	3.4519E-01
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XS ₀ COEF	0.787F-03
(10) CONSTANT	0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.2958E-04

*REFLECTIVE SURFACE NO. 6 IDENTIFICATION: LOWER FRONT PANEL RIGHT SIDE

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.3314E-01
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XS ₀ COEF	0.787F-03
(10) CONSTANT	0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.2958E-04

*REFLECTIVE SURFACE NO. 7 IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.0
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XS ₀ COEF	0.0
(10) CONSTANT	-0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.2322E-04

*REFLECTIVE SURFACE NO. 8 IDENTIFICATION: FORWARD SIDE PANEL RIGHT SIDE

VALUES FOR COEFFICIENTS OF SURFACE	
(1) X COEF	0.0
(4) XY COEF	0.0
(6) XZ COEF	0.0
(7) XS ₀ COEF	0.0
(10) CONSTANT	-0.1000E 01

ROOT MEAN SQUARE ERROR OF FIT= 0.2322E-04

B24

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** RESULTS OF CURVE-FIT ANALYSIS ***

BAREFLECTIVE SURFACE DATA

•REFLECTIVE SURFACE NO. 9

IDENTIFICATION: AFT SIDE PANEL LEFT SIDE

VALUES FOR COEFFICIENTS OF SURFACE			
		(2) Y COFF	(3) Z COFF
(1)	X COFF	0.0	0.0
(4)	XY COFF	0.0	0.0
(7)	XSQ COFF	0.0	0.0
(10)	CONSTANT	-0.1000E 01	0.4410E-03

ROOT MEAN SQUARE ERROR OF FIT= 0.4186E-15

•REFLECTIVE SURFACE NO. 10

IDENTIFICATION: AFT SIDE PANEL RIGHT SIDE

VALUES FOR COEFFICIENTS OF SURFACE			
		(2) Y COFF	(3) Z COFF
(1)	X COFF	0.0	0.0
(4)	XY COFF	0.0	0.0
(7)	XSQ COFF	0.0	0.0
(10)	CONSTANT	-0.1000E 01	0.4410E-03

ROOT MEAN SQUARE ERROR OF FIT= 0.4186E-15

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** SUN CLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 1

SUN ELEVATION: 0.0
 A/C PITCH ATTITUDE: 0.0
 A/C ROLL ATTITUDE: 0.0

IDENTIFICATION: WINDSHIELD FORWARD SECTION LEFT SIDE

XFCR 114.00
 YRCR C.C
 ZROT 1P.00
 DISTG 1500.00

A/C YAW

BOUNDARY POINTS -- BODY AXES

(STATIONLINE,PUTTLINE,WATERLINE)

X Y Z

			STATUS FLAGS FOR POINTS			
			TRFL ISHAD INTFL		AZIMUTH CLINT	ELEVATION
-45.00	48.00	-48.00	24.00	1	0	0
-45.00	16.25	-36.00	24.00	1	0	0
-45.00	6.43	-24.00	24.00	0	0	0
-45.00	0.0	0.0	24.00	0	0	-149.65
-45.00	6.43	0.0	48.00	0	0	-95.74
-45.00	16.25	6.0	60.00	0	0	-104.13
-45.00	48.00	0.0	72.00	0	0	-141.81
-45.00	48.00	0.0	72.00	0	0	-179.85
-45.00	48.00	-27.97	54.00	1	0	122.73
-45.00	48.00	-48.00	24.00	0	0	-99.75
-45.00	16.25	-76.00	24.00	0	0	-7.00
-45.00	6.43	-24.00	24.00	0	0	50.02
-45.00	0.0	0.0	24.00	0	0	89.74
-45.00	6.43	0.0	48.00	0	0	104.13
-45.00	16.25	0.0	60.00	0	0	141.81
-45.00	48.00	0.0	72.00	0	0	179.85
-45.00	48.00	-37.97	54.00	0	0	-122.51

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 2

*** SUN CLINT SIGNATURE ***

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

IDENTIFICATION: WINDSHIELD FORWARD SECTION RIGHT SIDE

XROT 114.00
 YROT 0.0
 ZROT 18.00
 DISTG 1500.00

A/C YAW

BOUNDARY POINTS -- BODY AXFS
 (STATIONLINE, BOTTLINE, WATERLINE)
 X Y

	X	Y	Z	STATUS	FLAGS FOR POINTS	CLINT	ELEVATION
-45.00	48.00	48.00	24.00	0	0	0	89.75
-45.00	16.25	36.00	24.00	0	0	0	7.00
-45.00	6.43	24.00	24.00	0	0	0	-30.02
-45.00	0.0	0.0	24.00	0	0	0	-89.74
-45.00	6.43	0.7	48.00	0	0	0	-108.13
-45.00	16.25	0.0	60.00	0	0	0	-141.81
-45.00	48.00	0.0	72.00	0	0	0	-179.85
-45.00	48.00	37.47	54.00	0	0	0	122.51
-45.00	48.00	48.00	24.00	1	0	0	43.77
-45.00	16.25	36.00	24.00	1	0	0	-89.95
-45.00	6.43	24.00	24.00	0	0	0	-173.10
-45.00	0.0	0.0	24.00	0	0	0	149.65
-45.00	45.00	0.0	24.00	0	0	0	89.74
-45.00	6.43	0.0	48.00	0	0	0	108.13
-45.00	16.25	0.0	60.00	0	0	0	141.81
-45.00	48.00	0.0	72.00	0	0	0	179.85
-45.00	48.00	37.47	54.00	1	0	0	0.17
-45.00	48.00	48.00	24.00	0	0	0	-43.69

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 3 IDENTIFICATION: WINDSHIELD AFT SECTION LEFT SIDE
 SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

A/C YAW	BOUNDARY POINTS -- BODY AXES			STATUS FLAGS FOR POINTS		
	X	Y	Z	INPL	ISHAD	INTRF
-45.00	72.00	0.0	72.00	0	0	0
-45.00	66.00	-31.75	60.00	1	0	0
-45.00	60.00	-41.57	48.00	1	0	0
-45.00	48.00	-48.00	24.00	1	0	0
-45.00	48.00	-37.47	54.00	1	0	0
-45.00	48.00	0.0	72.00	0	0	0
45.00	72.00	0.0	72.00	0	0	0
45.00	66.00	-31.75	60.00	0	0	0
45.00	60.00	-41.57	48.00	0	0	0
45.00	48.00	-48.00	24.00	0	0	0
45.00	48.00	-37.47	54.00	0	0	0
45.00	48.00	0.0	72.00	0	0	0

XROT 114.00
 YROT 0.2
 ZROT 18.00
 DIST 1500.00

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 4

IDENTIFICATION: WINDSHIELD AFT SECTION RIGHT SIDE

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.0
A/C ROLL ATTITUDE	0.0

A/C YAW

XROT	114.00
YROT	0.0
ZROT	18.00
DISTG	1500.00

POUNDRY POINTS -- BODY AXES

(STATIONLINE,BUTTLINE,WATERL INFR)

GLINT

AZIMUTH ELEVATION:

X	Y	Z	STATUS FLAG FOR POINTS	INRFL	ISNAU	INTRF
-45.00	72.00	0.0	72.00	0	0	0
-45.00	66.00	31.75	60.00	0	0	0
-45.00	60.00	41.57	48.00	0	0	0
-45.00	48.00	48.00	24.00	0	0	0
-45.00	48.00	37.47	54.00	0	0	0
-45.00	48.00	0.0	72.00	0	0	0
-45.00	72.00	0.0	72.00	0	0	0
-45.00	66.00	31.75	60.00	1	0	0
-45.00	60.00	41.57	48.00	1	0	0
-45.00	48.00	48.00	24.00	1	0	0
-45.00	48.00	37.47	54.00	1	0	0
-45.00	48.00	0.0	72.00	1	0	0
-45.00	72.00	0.0	72.00	1	0	0
-45.00	66.00	31.75	60.00	0	0	0
-45.00	60.00	41.57	48.00	0	0	0
-45.00	48.00	48.00	24.00	0	0	0
-45.00	48.00	37.47	54.00	0	0	0
-45.00	48.00	0.0	72.00	0	0	0
-45.00	72.00	0.0	72.00	0	0	0

CURVED PANELS WITH NO FENCES

SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 5

IDENTIFICATION: LOWER FRONT PANEL LEFT SIDE

A/C ELEVATION 0.0

A/C PITCH ATTITUDE 0.0

A/C ROLL ATTITUDE 0.0

XROT 114.00
 YROT 0.0
 ZROT 18.00
 DISTG 1500.00

A/C YAW	BOUNDARY POINTS -- PONY AXES			STATUS FLAGS FOR POINTS			GLINT AZIMUTH	ELLEVATION
	X	Y	Z	INRFL	ISHAD	INTRF		
-45.00	48.00	-41.57	0.0	1	0	0	108.44	37.84
-45.00	30.00	-37.47	0.0	1	0	0	150.08	16.68
-45.00	8.20	-12.00	0.0	0	0	0	-136.41	-24.21
-45.00	18.60	-12.00	-12.00	0	0	0	-159.91	-22.72
-45.00	48.00	-12.00	-22.48	1	0	0	176.15	19.96
-45.00	48.00	-37.47	-6.00	1	0	0	122.71	43.70
-45.00	48.00	-41.57	0.0	0	0	0	-108.22	-37.85
-45.00	30.00	-37.47	0.0	0	0	0	-54.12	-54.77
-45.00	8.20	-12.00	0.0	0	0	0	75.03	-49.61
-45.00	18.60	-12.00	-12.00	0	0	0	129.00	-6.18
-45.00	48.00	-12.00	-22.48	0	0	0	-176.25	-20.19
-45.00	48.00	-37.47	-6.00	0	0	0	-122.49	-43.75

CURVED PANELS WITH NO FERGES

SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. 6 IDENTIFICATION: LOWER FRONT PANFL RIGHT SIDE

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.0
A/C ROLL ATTITUDE	0.0

A/C YAW	BOUNDARY POINTS -- BODY AXES			STATUS FLAGS FOR POINTS			
	X	Y	Z	STATIONLINE,BOTTLINE,WATERLINE	TMFL	ISHAD	INTRF
-95.00	48.00	41.57	0.0	0	0	0	0
-95.00	30.00	37.47	0.0	0	3	0	54.12
-95.00	8.20	12.00	0.0	0	0	0	-75.03
-95.00	18.60	12.00	-12.00	0	0	0	-129.00
-95.00	48.70	12.00	-22.48	0	0	0	176.29
-95.00	48.00	37.47	-6.00	0	0	0	122.49
-95.00	48.00	41.57	0.0	1	0	0	-108.44
-95.00	30.00	37.47	0.0	1	0	0	-150.88
-95.00	8.20	12.00	0.0	0	0	0	136.41
-95.00	18.60	12.00	-12.00	0	0	0	159.91
-95.00	48.00	12.00	-22.48	1	0	0	-176.35
-95.00	48.00	37.47	-6.00	1	0	0	-122.71

CURVED PANELS WITH NO FENCES

REFLECTIVE SURFACE NO. 7

SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***
IDENTIFICATION: FORWARD SIDE PANEL LEFT SIDE

SUN ELEVATION 0.0
 A/C PITCH ATTITUDE 0.0
 A/C ROLL ATTITUDE 0.0

A/C YAW

BOUNDARY POINTS -- POLY ARFS
(STATIONLINE•BOTTLINE•WATERLINE)
X Y Z

			STATUS FLAGS FOR POINTS			
			INRFL	ISHAD	INTRF	CLINT
						AZIMUTH ELEVATION
-45.00	72.00	-48.00	24.00	1	0	90.01 0.05
-45.00	96.00	-48.00	24.00	1	0	90.07 0.05
-45.00	96.00	-44.50	42.00	1	0	99.37 -29.44
-45.00	96.00	-31.75	60.00	1	0	142.26 -44.50
-45.00	84.00	-31.75	60.00	1	0	142.26 -44.52
-45.00	72.00	-46.48	36.00	1	0	53.83 -20.02
-45.00	72.00	-48.00	24.00	0	0	-89.80 -0.01
-45.00	96.00	-48.00	24.00	0	0	-89.85 -0.01
-45.00	96.00	-44.50	42.00	0	0	-99.14 29.44
-45.00	96.00	-31.75	60.00	0	0	-142.05 44.65
-45.00	84.00	-31.75	60.00	0	0	-142.06 44.67
-45.00	72.00	-46.48	36.00	0	0	-93.61 20.01

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE NO. K

IDENTIFICATION: FORWARD SIDE PANEL RIGHT SIDE

SUN ELEVATION 0.0

A/C PITCH ATTITUDE 3.5

A/C ROLL ATTITUDE 0.0

A/C YAW	REFLECTION POINTS -- BODY AXES (STATIONLINE, RULLINE, WATERLINE)	STATUS FLAGS FOR POINTS ISLAND INTR	GLINT AZIMUTH ELEVATION
-45.00	72.00 48.00 24.00	C 0 0 r	A9.80 -C.01
-45.00	96.00 48.00 24.00	C 0 0 C	A9.85 -0.01
-45.00	96.00 48.50 42.00	0 0 0	99.14 29.49
-45.00	96.00 31.75 60.00	0 0 0	142.05 44.65
-45.00	84.00 31.75 60.00	0 0 0	142.06 44.67
-45.00	72.30 46.48 36.00	0 0 0	93.61 20.36
-45.00	72.00 48.00 24.00	1 0 0	-90.01 0.05
-45.00	96.00 48.00 24.00	1 0 0	-90.07 0.05
-45.00	96.00 44.50 42.00	1 0 0	-99.37 -29.44
-45.00	96.00 31.75 60.00	1 0 0	-142.26 -44.50
-45.00	64.00 31.75 60.00	1 0 0	-142.26 -44.52
-45.00	72.00 46.48 36.00	1 0 0	-93.63 -20.32

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

*** SUN GLINT SIGNATURE ***

REFLECTIVE SURFACE AC. 9

IDENTIFICATION: AFT SIDE PANEL LEFT SIDE

SUN ELEVATION	0.0
A/C PITCH ATTITUDE	0.6
A/C ROLL ATTITUDE	0.0

A/C YAW

BOUNDARY POINTS -- PONY AXES

(STATIONLINE,BUTTLINE,WAFFLINE)

X	Y	Z	STATUS FLAGS FOR POINTS	TYPFL	TSNAD	TINRF	GLINT	AZIMUTH	ELEVATION
108.00	-48.00	24.00		1	0	0	90.009	0.004	
144.00	-48.00	24.00		1	0	0	90.018	0.004	
144.00	-44.50	42.00		1	0	0	99.446	-29.359	
144.00	-31.75	60.00		1	0	0	142.23	-44.43	
108.00	-31.75	60.00		1	0	0	142.24	-44.49	
108.00	-44.50	42.00		1	0	0	99.440	-29.43	
108.00	-48.00	24.00		0	0	0	-89.88	-0.00	
144.00	-48.00	24.00		0	0	0	-89.86	-0.70	
144.00	-44.50	42.00		0	0	0	-99.23	29.43	
144.00	-31.75	60.00		0	0	0	-142.03	44.58	
108.00	-31.75	60.00		0	0	0	-142.04	44.63	
108.00	-44.50	42.00		0	0	0	-99.17	29.47	
45.00									

CURVED PANELS WITH NO FENCES SAMPLE CASE TWO

REFLECTIVE SURFACE NO. 10 IDENTIFICATION: AFT SIDE PANEL RIGHT SIDE
*** SUN GLINT SIGNATURE ***

SUN ELEVATION 0.0

A/C PITCH ATTITUDE 0.0

A/C ROLL ATTITUDE 0.0

A/C YAW

X	Y	Z	BOUNDARY POINTS - BODY AXES (STATIONLINE, RUTLINE, WATERLINE)			AZIMUTH	GLINT ELEVATION
			INFL	ISRF	ISHAD		
106.00	46.00	24.00	0	0	0	-9.98	-0.00
144.00	46.00	24.00	0	0	0	89.96	-6.00
144.00	44.50	42.00	0	0	0	59.23	29.43
144.00	31.75	60.00	0	0	0	142.03	44.58
108.00	31.75	60.00	0	0	0	142.04	44.63
108.00	44.50	42.00	0	0	0	99.17	29.47
108.00	46.00	24.00	1	0	0	-90.09	0.04
144.00	46.00	24.00	1	0	0	-90.16	0.04
144.00	44.50	42.00	1	0	0	-99.46	-29.39
144.00	31.75	60.00	1	0	0	-142.23	-44.43
108.00	31.75	60.00	1	0	0	-142.24	-44.49
128.00	44.50	42.00	1	0	0	-99.40	-29.43

APPENDIX B
PROGRAM LISTING

G LEVEL 21	MAIN	DATE = 78103	08/54/44
***** PROGRAMMED BY FRED WHITE ON BOEING IBM 370 *****			
***** BOEING VERTOL, PHILADELPHIA, PENNA. *****			
***** AREA CODE 215 522-2256 *****			
C			00000043
C			00000044
REAL*8 XCURE(450),FCURF(50),SIG			00000045
REAL*8 CFALP(9)			00000050
DIMENSION ALP(50),BALP(50)			00000060
DIMENSION SIGF(21),SIGS(12)			00000070
DIMENSION COEFFR(80),COEFSE(120)			00000080
DIMENSION DATAAC(30),DATAAF(200),DATAS(1500),TITLES(960),			00000090
1 MF(20),MS(121),DWORKS(1201),WKSP2(120)			00000100
DIMENSION ALPHA2(800),BETA2(800),INTRFL(800),ISHADW(800),			00000110
1 INTERF(800)			00000120
DIMENSION IPTL0L(432)			00000130
DIMENSION IBUF(1000),XPLOT(63),YPLOT(63),APLOT(432),BPLOT(432)			00000140
COMMON/TOLRS/TOLEQN,TOLSNT,TOLQPT,TOLINT,TOLCK1,TOLCK2,			00000150
1 TOLCK4,DELT,IOLCIR,ANGST2,TOLPLT			00000160
TOLEQN=0.00001			00000170
TOLSNT=0.0001			00000171
TOLQPI=0.0010001			00000172
TOLINT=0.02			00000173
TOLCK1=0.02			00000174
TOLCK2=0.02			00000175
TOLCK4=0.02			00000176
DELT=0.10			00000177
IOLCIR=0.0001			00000178
ANGST2=90.			00000179
TOLPLT=0.0001			00000180
DO 5 I=1,30			00000181
5 DATAAC(I)=0.0			00000182
DO 6 I=1,200			00000190
6 DATAAF(I)=0.			00000200
DO 7 I=1,1500			00000210
7 DATAS(I)=0.			00000220
CALL INPUT(DATAAC,DATAAF,DATAS,TITLES,MF,MS,0)			00000230
KPRINT=DATAAC(2)+0.5			00000240
IPLOT=DATAAC(24)+0.5			00000250
IPROBLE=DATAAC(25)+0.5			00000260
CALL OUTPT1(DATAAC,DATAAF,DATAS,TITLES,MF,MS)			00000270
CALL INPUT(DATAAC,DATAAF,DATAS,TITLES,MF,MS+1)			00000280
NFENCE=DATAAC(22)+0.5			00000290
IF(NFENCE.EQ.0) GO TO 14			00000300
DO 12 I=1,NFENCE			00000310
L=MF(1)			00000320
NPTS=DATAAF(L)+0.5			00000330
ISTART=L+1			00000340
CALL CFITE(DATAAF,ISTART,NPTS,COEFFR,I,SIG,XCURE,FCURF)			00000350
SIGF(I)=SIG			00000360
12 CONTINUE			00000370
IF(KPRINT.LT.2) GO TO 14			00000380
CALL OUTPT2(COEFFR,SIGF,NFENCE,TITLES,1)			00000390
			00000400

G LEVEL	21	MAIN	DATE = 78103	08/54/44
14	CONTINUE		00000410	
	NPANEL=DATAAC(23)+0.5		00000420	
DO	15 I=1, NPANEL		00000430	
	L=MS(I)		00000440	
	NPTS=DATAS(L)+0.5		00000450	
	ISTART=L+1		00000460	
	CALL CFITS(DATAAC,DATAS,ISTART,NPTS,COEFSB,I,SIG,XCURF,FCURF)		00000470	
	SIGS(I)=SIG		00000480	
15	CONTINUE		00000490	
	IF(KPRINT.LT.2) GO TO 16		00000500	
	CALL OUTPT2(COEFSB,SIGS,NPANEL,TITLES,2)		00000510	
16	CONTINUE		00000520	
	NGAM=DATAAC(6)+0.5		00000530	
	DGAM=DATAAC(8)		00000540	
	GAMMA=DATAAC(7)-DGAM		00000550	
	IF(IPLOT.EQ.0.OR.IPLCT.EQ.2) GO TO 17		00000560	
	JCONT=0		00000570	
	CALL PLOTS(IHUF,1000,9)		00000580	
17	CONTINUE		00000590	
DO	30 I=1, NGAM		00000600	
	GAMMA=GAMMA+DGAM		00000610	
	CALL GLINT(DATAAC,DATAS,MS,COEFSB,GAMMA,ALPHA2,BETA2,INTRFL,		00000620	
1	ISHADW,INTERF,DWORKS,DATAF,MF)		00000630	
	IF(KPRINT.LT.1) GO TO 18		00000640	
	CALL OUTPT3(DATAAC,DATAS,MS,TITLES,GAMMA,ALPHA2,BETA2,INTRFL,		00000650	
	1 ISHADW,INTERF)		00000660	
18	CONTINUE		00000670	
	CALL MINMAX(DATAAC,DATAS,MS,ALPHA2,BETA2,APOINT,BPOINT,INTRFL,		00000680	
1	ISHADW,INTERF,IPLTOL)		00000690	
	IF(IPROBL.EQ.0) GO TO 19		00000700	
	CALL PROPL(DATAAC,DATAS,MS,APOINT,BPOINT,XPOINT,YPOINT,DWORKS,DWKSP2,		00000710	
1	VPROBL,IPLTOL)		00000720	
19	CONTINUE		00000730	
	IF(IPLOT.EQ.0.OR.IPLOT.EQ.1) GO TO 20		00000740	
	CALL PLOTOL(DATAAC,DATAS,MS,TITLES,GAMMA,APOINT,BPOINT,		00000750	
1	VPROBL,IPLTOL)		00000760	
20	CONTINUE		00000770	
	IF(IPLOT.EQ.0.OR.IPLOT.EQ.2) GO TO 21		00000780	
	CALL DRPLOT1(DATAAC,DATAS,MS,GAMMA,ALPHA2,BETA2,INTRFL,		00000790	
1	ISHADW,INTERF,XPOINT,YPOINT,IFUF,JCONT,DWORKS,DWKSP2,VPROBL)		00000800	
21	CONTINUE		00000810	
30	CONTINUE		00000820	
	IF(IPLOT.EQ.0.OR.IPLOT.EQ.2) GO TO 31		00000830	
	CALL PLOT(10.0,0,999)		00000840	
31	CONTINUE		00000850	
	STOP		00000860	
	END		00000870	

G LEVEL	21	INPUT	DATE = 78102	10/47/48
		SUBROUTINE INPUT(DATAAC,DATAF,DATA\$+TITLES,MF,MS,IDLTRAN)	00000880	
		DIMENSION DATAAC(1),DATAF(1),DATA\$(1),TITLES(1),MF(1),MS(1)	00000890	
		DATA BLANK/1H /	00000900	
		IF(IDLTRAN.EQ.1) GO TO 80	00000930	
		READ(5,101) (TITLES(I),I=1,72)	00000940	
		DO 2 I=73,80	00000950	
2		TITLES(I)=BLANK	00000960	
		LT=80	00000970	
		READ(5,100) DEFLT,PRINT,THETA,PHI,	00001000	
		IF(DEFLT.LT.0.5) DEFLT=0.	00001010	
		IF(PRINT.LT.0.5) PRINT=0.	00001020	
		IF(ABS(THETA).LT.0.001) THETA=0.	00001030	
		IF(ABS(PHI).LT.0.001) PHI=0.	00001040	
		DATAAC(1)=DEFLT	00001050	
		DATAAC(2)=PRINT	00001060	
		DATAAC(3)=0.	00001070	
		DATAAC(4)=THETA	00001080	
		DATAAC(5)=PHI	00001090	
		READ(5,100) GAMN,GAMI,DGAM,PSIN,PSII,DPSI	00001100	
		IF(GAMN.GT.0.5) GO TO 3	00001110	
		GAMN=3.	00001120	
		GAMI=0.	00001130	
		DGAM=30.	00001140	
3		IF(PSIN.GT.0.5) GO TO 5	00001150	
		PSIN=9.	00001160	
		PSII=-180.	00001170	
		DPSI=45.	00001180	
5		DATAAC(6)=GAMN	00001190	
		DATAAC(7)=GAMI	00001200	
		DATAAC(8)=DGAM	00001210	
		DATAAC(9)=PSIN	00001220	
		DATAAC(10)=PSII	00001230	
		DATAAC(11)=DPSI	00001240	
		READ(5,100) XROT,YROT,ZROT	00001250	
		IF(ABS(XROT).LT.0.001) XROT=0.	00001260	
		IF(ABS(YROT).LT.0.001) YROT=0.	00001270	
		IF(ABS(ZROT).LT.0.001) ZROT=0.	00001280	
		DATAAC(12)=XROT	00001290	
		DATAAC(13)=YROT	00001300	
		DATAAC(14)=ZROT	00001310	
		READ(5,100) DISTG,ELVMIN,ELVMAX,AZMMIN,AZMMAX	00001320	
		IF(ABS(DISTG).LT.0.001) DISTG=1000.	00001330	
		IF(ABS(ELVMIN).GT.0.001) GO TO 6	00001340	
		IF(ABS(ELVMAX).GT.0.001) GO TO 6	00001350	
		ELVMIN=-10.	00001360	
		ELVMAX=10.	00001370	
6		IF(ABS(AZMMIN).GT.0.001) GO TO 7	00001380	
		IF(ABS(AZMMAX).GT.0.001) GO TO 7	00001390	
		AZMMIN=-180.	00001400	
		AZMMAX=180.	00001410	
7		DATAAC(15)=DISTG	00001420	
		DATAAC(16)=ELVMIN	00001430	

G LEVEL	21	INPUT	DATE = 78102	10/47/48
		DATA(17)=ELVMAX		00001440
		DATA(18)=AZMMIN		00001450
		DATA(19)=AZMMAX		00001460
		READ(5,100) FENCES,PANELS,PLOT,PROBL		00001470
		DATA(22)=FCNCES		00001480
		DATA(23)=PANELS		00001490
		DATA(24)=PLOT		00001500
		DATA(25)=PROBL		00001510
		LOC=1		00001520
		IF(DATA(22).LT.0.5) GO TO 30		00001530
		NFENCE=DATA(22)+0.5		00001540
	DO 20 I=1,NFENCE			00001550
		READ(5,100) DATAF(LOC)		00001560
		NPTS=DATAF(LOC)+0.5		00001570
		MFC(I)=LOC		00001580
		N3=3*NPTS		00001590
		READ(5,100) (DATAF(LOC+J),J=1,N3)		00001600
20		LOC=LOC+N3+1		00001610
	30	NPANEL=DATA(23)+0.5		00001620
		LOC=1		00001630
	DO 40 I=1,NPANEL			00001640
		MS(I)=LOC+20		00001650
		READ(5,101) (TITLES(LT+J),J=1,72)		00001660
	DO 301 J=73,80			00001670
301		TITLES(LT+J)=BLANK		00001680
		LT=LT+80		00001690
		READ(5,100) DATAS(LOC)		00001700
		IF(DATAS(LOC).GT.0.5) GO TO 32		00001710
		IF(DATA(1).GT.0.5) GO TO 310		00001720
		READ(5,100) (DATAS(LOC+J),J=1,9)		00001730
		READ(5,100) (DATAS(LOC+J),J=10,19)		00001740
310		READ(5,100) DATAS(LOC+20)		00001750
		NPTS=DATAS(LOC+20)+0.5		00001760
		N3P20=20+3*NPTS		00001770
		READ(5,100) (DATAS(LOC+J),J=21,N3P20)		00001780
	GO TO 38			00001790
32		IF(I.EQ.1) GO TO 70		00001840
		ISYM=DATAS(LOC)+0.5		00001850
	DO 34 J=1,20			00001860
34		DATAS(LOC+J)=DATAS(LOC+J)		00001870
		NPTS=DATAS(LOC+20)+0.5		00001880
	DO 35 J=1,NPTS			00001890
		L=(J-1)*3		00001900
		DATAS(LOC+21+L)=DATAS(LOC+21+L)		00001910
		DATAS(LOC+22+L)=DATAS(LOC+22+L)		00001920
		DATAS(LOC+23+L)=DATAS(LOC+23+L)		00001930
		IF(ISYM.EQ.1) DATAS(LOC+22+L)=-DATAS(LOC+22+L)		00001940
		IF(ISYM.EQ.2) DATAS(LOC+23+L)=-DATAS(LOC+23+L)		00001950
		IF(ISYM.EQ.3) DATAS(LOC+21+L)=-DATAS(LOC+21+L)		00001960
35		CONTINUE		00001970
38		LOCP=LOC		00001980
40		LOC=LOC+21+3*NPTS		00001990

G LEVEL	21	INPUT	DATE = 78102	10/47/48
		NT=N PANEL +1	00002000	
		DO 68 I=1,NT	00002010	
		L=(I-1)*80	00002020	
		NB=0	00002030	
		DO 62 J=1,80	00002040	
		IF(TITLESL+J).NE.BLANK) GO TO 63	00002050	
62		NB=N B+1	00002060	
63		IF(NB.EQ.80) GO TO 68	00002070	
		IF(NB.EQ.0) GO TO 68	00002080	
		DO 66 J=1,80	00002090	
		K=J+N B	00002100	
		IF(K.GT.80) GO TO 64	00002110	
		TITLESL+J)=TITLESL+K)	00002120	
		GO TO 66	00002130	
64		TITLESL+J)=BLANK	00002140	
66		CONTINUE	00002150	
68		CONTINUE	00002160	
		RETURN	00002170	
70		WRITE(6,500)	00002180	
		STOP	00002190	
500		FORMAT(1H1,*ERROR IN INPUT. SYMMETRICAL PANEL CANNOT BE FIRST*)	00002200	
100		FORMAT(6E10.0)	00002210	
101		FORMAT(80A1)	00002220	
80		CONTINUE	00002230	
		IF(DATAC(22).LT.0.5) GO TO 90	00002240	
		NFENCE=DATAC(22)+0.5	00002250	
		DO 86 I=1,NFENCE	00002260	
		LCC=MF(I)	00002270	
		NPTS=DATAF(LCC)+0.5	00002280	
		DO 86 J=1,NPTS	00002290	
		DO 86 JJ=1,3	00002300	
		LJ=(J-1)*3+JJ	00002310	
86		DATAF(LOC+LJ)=DATAF(LOC+LJ)-DATAC(JJ+11)	00002320	
88		CONTINUE	00002330	
90		CONTINUE	00002340	
		N P A N E L = D A T A C (2 3) + 0 . 5	00002350	
		DO 96 I=1,N P A N E L	00002360	
		L O C P 2 0 = M S (I)	00002370	
		N P T S = D A T A S (L O C P 2 0) + 0 . 5	00002380	
		DO 94 J=1,N P T S	00002390	
		DO 94 JJ=1,3	00002400	
		LJ=(J-1)*3+JJ	00002401	
94		DATAS(LOC P20+LJ)=DATAS(LOC P20+LJ)-DATAC(JJ+11)	00002410	
96		CONTINUE	00002420	
		RETURN	00002430	
		END	00002440	

G LEVEL	21	OUTPT1	DATE = 78102	10/47/48
		SUBROUTINE OUTPT1(DATAC,DATAF,DATAS,TITLES,MF,MS)		00002450
		DIMENSION DATAC(1),DATAF(1),DATAS(1),TITLES(1),MF(1),MS(1)		00002460
		WRITE(6,102) (TITLES(I),I=1,80)		00002470
102		FORMAT(1H1*80A1)		00002480
		WRITE(6,101)		00002490
101		FORMAT(1H0,51X,*++ INPUT DATA ++*)		00002500
		WRITE(6,103) DATAC(1),DATAC(2),DATAC(24),DATAC(25)		00002510
103		FORMAT(1H0,2X,*++ CONTROL OPTIONS*/1H .5X,*DEFAULT OPTION*,7X,F2.0,0,00002520 1 8X,*PRINT OPTION*,9X,F2.0/		00002530
		2 1H .5X,*PLOT OPTION*,10X,F2.0,8X,*PROBABILITY OPT*,6X,		00002540
		3 F2.0)		00002550
		WRITE(6,104) DATAC(4),DATAC(5),DATAC(10)		00002560
104		FORMAT(1H0,2X,*++ AIRCRAFT INITIAL EULER ORIENTATION*/1H .5X,*PITCH 00002570 1*,13X,F6.1,7X,*ROLL*,14X,F6.1,7X,*YAW*,15X,F6.1)		00002580
		WRITE(6,105) (DATAC(I),I=9,11)		00002590
105		FORMAT(1H0,2X,*++ SELECTED YAW ANGLE ROTATIONS*/1H .5X,*NO. OF ANGL 00002600 2ES*,6X,F4.0,8X,*INITIAL ANGLE*,5X,F6.1,7X,*ANGLE INCREMENT*,3X,F6.0,00002610 31)		00002620
		WRITE(6,106) (DATAC(I),I=6,8)		00002630
106		FORMAT(1H0,2X,*++ SELECTED SUN ELEVATIONS*/1H .5X,*NO. OF ANGLES*, 1 6X,F4.0,8X,*INITIAL ANGLE*,5X,F6.1,7X,*ANGLE INCREMENT*,3X,F6.1)		00002640
		WRITE(6,107) DATAC(15),DATAC(12),DATAC(13),DATAC(14)		00002650
107		FORMAT(1H0,2X,*++ REFERENCE INFORMATION FOR MEASURING SUN GLINT ANG 00002670 1LES*/1H .5X,*REFERENCE POINT FOR MEASURING ANGLES OF GLINT SIGNATU 00002680		
		2RE*,5X,*DISTANCE AT WHICH GLINT SIGNATURE FORMED (FT.)*,F10.3/1H .00002690		
		3 18X,*X*,9X,*Y*,9X,*Z*/1H .14X,3(F7-2,3X))		00002700
		WRITE(6,108) DATAC(18),DATAC(16),DATAC(19),DATAC(17)		00002710
108		FORMAT(1H0,2X,*++ PLOT SCALING*/1H .5X,*GLINT AZIMUTH*,18X,*GLINT E 00002720 . ILEVATION*/1H .7X,*MINIMUM*,8X,F6.1,10X,*MINIMUM*,8X,F6.1/1H .7X, 2 *MAXIMUM*,8X,F6.1,10X,*MAXIMUM*,8X,F6.1)		00002730
		WRITE(6,109) DATAC(18),DATAC(16),DATAC(19),DATAC(17)		00002750
109		FORMAT(1H0,2X,*++ BOUNDARIES FOR CALCULATING PROBABILITY*/1H .5X, 1 *GLINT AZIMUTH*,18X,*GLINT ELEVATION*/1H .7X,*MINIMUM*,8X,F6.1, 2 10X,*MINIMUM*,8X,F6.1/1H .7X,*MAXIMUM*,8X,F6.1,10X,*MAXIMUM*,8X, 3F6.1)		00002770
		IF(DATAC(22).LT.0.5) GO TO 25		00002790
		WRITE(6,102) (TITLES(MMM),MMM=1,80)		00002810
		WRITE(6,101)		00002820
		WRITE(6,110) DATAC(22)		00002830
110		FORMAT(1H0,2X,*++ FENCE DATA*/1H .5X,*NO. OF FENCES*,7X,F3.0)		00002840
		LINES=5		00002850
		NFENCE=DATAC(22)+0.5		00002860
		DO 20 I=1,NFENCE		00002870
		M2=MF(I)		00002880
		WRITE(6,111)		00002890
		LINES=LINES+2		00002900
		IF((LINES*6).LE.50) GO TO 4		00002910
		WRITE(6,102) (TITLES(MMM),MMM=1,80)		00002920
		WRITE(6,101)		00002930
		WRITE(6,111)		00002940
		LINES=5		00002950
4		WRITE(6,112) I,DATAF(M2)		00002960

G LEVEL 21	OUTPT1	DATE = 78102	10/47/48
111	FORMAT(1H0)		00002970
112	FORMAT(1H ,5X, "FENCE NO.", I3/1H ,7X, "NUMBER OF POINTS", 4X, F3.0/		00002980
1	1H0,7X, "BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE,WATERLINE)		00002990
	XLINE)*		00003000
1		/1H ,3(18X, "X", 9X, "Y", 9X,	00003010
2	"Z", 2X))		00003020
	LINES=LINES+5		00003030
	NPTS=DATAF(M2)+0.5		00003040
	M03=NPTS/3		00003050
DO 10	L=1+ND3		00003060
J=(L-1)*9+M2			00003070
K=(L-1)*3+1			00003080
KP1=K+1			00003090
KP2=K+2			00003100
IF((LINES+1).LE.50) GO TO 8			00003110
WRITE(6,102) (TITLES(MMM),MMM=1,80)			00003120
WRITE(6,101)			00003130
WRITE(6,113) I			00003140
113	FORMAT(1H0,5X, "FENCE NO.", I3/1H ,7X, "BOUNDARY POINTS -- BODY AXES		00003150
X (STATIONLINE,BUTTLINE,WATERLINE)*			00003160
1 /1H ,3(18X, "X", 9X, "Y", 9X, "Z", 2X))			00003170
	LINES=7		00003180
8 WRITE(6,114) K,(DATAF(J+M),M=1,3),KP1,(DATAF(J+M),M=4,6),			00003190
1 KP2,(DATAF(J+M),M=7,9)			00003200
114	FORMAT(1H ,2X,3(6X,"(",I2,")"+2X+2(F7.2,3X),F7.2,2X))		00003210
10 LINES=LINES+1			00003220
IF(J+ND3.EQ.NPTS) GO TO 20			00003230
J=ND3+9*M2			00003240
K=3+ND3+1			00003250
IF((J+ND3+1).EQ.NPTS) WRITE(6,115) K,(DATAF(J+M),M=1,3)			00003260
KP1=K+1			00003270
IF((3+ND3+2).EQ.NPTS) WRITE(6,116) K,(DATAF(J+M),M=1,3),			00003280
1 KP1,(DATAF(J+M),M=4,6)			00003290
	LINES=LINES+1		00003300
115	FORMAT(1H ,8X,"(",I2,")"+2X+2(F7.2,3X),F7.2)		00003310
116	FORMAT(1H ,2X+2(6X,"(",I2,")"+2X+2(F7.2,3X),F7.2,2X))		00003320
20 CONTINUE			00003330
25 CONTINUE			00003340
WRITE(6,102) (TITLES(MMM),MMM=1,80)			00003350
WRITE(6,101)			00003360
NPANEL=DATAF(23)+0.5			00003370
WRITE(6,117) DATAF(23)			00003380
117	FORMAT(1H0,2X,"**REFLECTIVE SURFACE DATA*/1H ,5X,"NO. OF SURFACES"		00003390
1,5X,F3.0)			00003400
	LINES=6		00003410
LT=0			00003420
DO 70 I=1,NPANEL			00003430
LT=LT+80			00003440
M2=MS(I)-20			00003450
IF((LINES+6).LE.50) GO TO 27			00003460
WRITE(6,102) (TITLES(MMM),MMM=1,80)			00003470
WRITE(6,101)			00003480

G LEVEL	21	OUTPT1	DATE = 78102	10/47/48
LINES=3				00003490
27 WRITE(6,118) I,(TITLES(LT+J),J=1,80)				00003500
118 FORMAT(1H0,5X,*REFLECTIVE SURFACE NO.,I3,5X,*IDENTIFICATION: *,				00003510
1 80A1)				00003520
LINES=LINES+2				00003530
IF((LINES+5).LE.50) GO TO 30				00003540
WRITE(6,102) (TITLES(MMM),MMM=1,80)				00003550
WRITE(6,101)				00003560
LINES=3				00003570
WRITE(6,118) I,(TITLES(LT+J),J=1,80)				00003580
LINES=LINES+2				00003590
30 WRITE(6,120) (DATAS(M2+J),J=1,9)				00003600
120 FORMAT(1H0,7X,*CURVE-FIT INPUT DATA*/1H ,9X,*DESIRED TERMS FOR CURE	00003610			
1VE-FITTING*/1H ,11X,*(1) X TERM *,F4.0,5X,*(2) Y TERM *,F4.0,	00003620			
2 5X,*(3) Z TERM *,F4.0,5X,*(4) XY TERM *,F4.0,5X,*(5) XZ TERM	00003630			
3*,F4.0/1H ,11X,*(6) YZ TERM *,F4.0,5X,*(7) XSQ TERM *,F4.0,5X,	00003640			
4 *(8) YSQ TERM *,F4.0,5X,*(9) ZSQ TERM *,F4.0)	00003650			
LINES=LINES+5				00003660
IF((LINES+6).LE.50) GO TO 40				00003670
WRITE(6,102) (TITLES(MMM),MMM=1,80)				00003680
WRITE(6,101)				00003690
LINES=3				00003700
WRITE(6,118) I,(TITLES(LT+J),J=1,80)				00003710
LINES=LINES+2				00003720
40 WRITE(6,121) (DATAS(M2+J),J=10,18)				00003730
121 FORMAT(1H0,9X,*INPUT VALUES FOR COEFFICIENTS OF SURFACE*/1H ,11X,	00003740			
1 *(1) X COEF *,E12.4,5X,*(2) Y COEF *,E12.4,5X,*(3) Z COEF *	00003750			
X*,E12.4/1H ,11X,				00003760
2 *(4) XY COEF *,E12.4,5X,*(5) XZ COEF *,E12.4,5X,*(6) YZ COEF *	00003770			
3,E12.4/1H ,11X,*(7) XSG COEF *,E12.4,5X,*(8) YSG COEF *,E12.4,5X,	00003780			
4 *(9) ZSQ COEF *,E12.4)				00003790
WRITE(6,123) DATAS(M2+19)				00003800
123 FORMAT(1H ,10X,*(10) CONSTANT *,E12.4)				00003810
LINES=LINES+6				00003820
IF((LINES+4).LE.50) GO TO 50				00003830
WRITE(6,102) (TITLES(MMM),MMM=1,80)				00003840
WRITE(6,101)				00003850
LINES=3				00003860
WRITE(6,118) I,(TITLES(LT+J),J=1,80)				00003870
LINES=LINES+2				00003880
50 WRITE(6,122)				00003890
122 FORMAT(1H0,7X,*BOUNDARY POINTS -- BODY AXES (STATIONLINE,BUTTLINE	00003900			
X,WATERLINE)*/1H ,3(18X,*X*,9X,				00003910
1 *Y*,9X,*Z*,2X))				00003920
LINES=LINES+3				00003930
NPTS=DATAS(M2+20)+0.5				00003940
ND3=NPTS/3				00003950
DO 60 L=1,ND3				00003960
J=(L-1)*9+20*M2				00003970
K=(L-1)*3+1				00003980
KP1=K+1				00003990
KP2=K+2				00004000

G LEVEL	21	OUTPT1	DATE = 78102	10/47/48
		IF((LINES+1).LE.50) GO TO 52		00004010
		WRITE(6,102) (TITLES(MMM),MMM=1,80)		00004020
		WRITE(6,101)		00004030
		WRITE(6,118) I,(TITLES(LT+K),K=1,80)		00004040
		LINES=5		00004050
52		WRITE(6,114) K,(DATAS(J+M),M=1,3),KP1,(DATAS(J+M),M=4,6),		00004060
		1 KP2,(DATAS(J+M),M=7,9)		00004070
60		60 LINES=LINES+1		00004080
		IF(3*ND3.EQ.NPTS) GO TO 70		00004090
		J=ND3*9+20*M2		00004100
		K=ND3*3+1		00004110
		IF((3*ND3+1).EQ.NPTS) WRITE(6,115) K,(DATAS(J+M),M=1,3)		00004120
		KP1=K+1		00004130
		IF((3*ND3+2).EQ.NPTS) WRITE(6,116) K,(DATAS(J+M),M=1,3),		00004140
		1 KP1,(DATAS(J+M),M=4,6)		00004150
		LINES=LINES+1		00004160
70		CONTINUE		00004170
		RETURN		00004180
		END		00004190

G LEVEL 21

CFITF

DATE = 78102

10/47/48

```
SUBROUTINE CFITF(DATAF,ISTART,NPTS,COEFFB,ICOF,SIG,XCURF,FCURF)      00004200
REAL*8 XCURF(1),FCURF(1),COFB(3),SIG,DET                           00004210
REAL*8 XPTS(10),YPTS(10),ZPTS(10)                                     00004211
DIMENSION IFIT(9)                                                 00004212
DIMENSION DATAF(1),COEFFB(1)                                         00004220
CALL NORM(DATAF,ISTART,NPTS,TNORM)                                    00004230
DO 10 L=1,NPTS                                                 00004250
LJ=(L-1)*3+ISTART-1                                              00004260
FCURF(L)=1.                                                       00004270
XPTS(L)=DATAF(LJ+1)/TNORM                                           00004271
YPTS(L)=DATAF(LJ+2)/TNORM                                           00004272
ZPTS(L)=DATAF(LJ+3)/TNORM                                           00004273
DO 10 J=1,3                                                       00004290
JJ=(L-1)*3+J                                                       00004300
10 XCURF(JJ)=DATAF(LJ+J)/TNORM                                     00004310
CALL CURFIT(FCURF,XCURF+3,NPTS,COFB,DET,SIG,1)                      00004330
DO 15 I=1,3                                                       00004331
15 IFIT(I)=1                                                       00004332
DO 18 I=4,9                                                       00004333
18 IFIT(I)=0                                                       00004334
CALL SIGNF(COFB,3,DET,XPTS,YPTS,ZPTS,NPTS,IFIT)                      00004340
L=(ICOF-1)*4                                                       00004350
DO 20 I=1,3                                                       00004360
20 COEFFB(L+I)=COFB(I)/TNORM                                       00004370
COEFFB(L+4)=-DET                                                 00004380
RETURN                                                       00004390
END                                                       00004400
```

G LEVEL 21

NORM

DATE = 78102

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SUBROUTINE NORM(DATA,ISTART,NPTS,TNORM)	00004410
DIMENSION DATA(1)	00004420
TNORM=0.	00004430
DO 20 I=1,NPTS	00004440
L=(I-1)*3+ISTART-1	00004450
DO 20 J=1,3	00004460
20 TNORM=TNORM+ DATA(L+J)**2	00004470
TNORM=TNORM/3./NPTS	00004480
TNORM=SQRT(TNORM)	00004490
RETURN	00004500
END	00004510

G LEVEL 21

CURFIT

DATE = 78102

10/47/48

```
SUBROUTINE CURFIT(F,X,N,M,COEF,DET,SIG,NH)          00004520
IMPLICIT REAL*8 (A-H,O-Z)                          00004530
DIMENSION A(9*9),B(9),X(1),F(1),COEF(1)           00004540
DO 10 I=1,N                                         00004550
B(I)=0.0                                           00004560
DO 10 J=1,N                                         00004570
10 A(I,J)=0.0
DO 20 I=1,M                                         00004580
DO 20 J=1,N                                         00004590
20 JJ=J*(I-1)*N                                     00004600
B(J)=B(J)+F(I)*X(JJ)
DO 20 L=1,N                                         00004610
LL=L*(I-1)*N                                     00004620
20 A(L,J)=A(L,J)+X(JJ)*X(LL)
CALL EQNSOL(A,B,N,COEF,DET,NH)                     00004630
SIG=0.0                                             00004640
DO 40 I=1,M                                         00004650
SUM=0.0                                           00004660
DO 30 J=1,N                                         00004670
JJ=J*(I-1)*N                                     00004680
30 SUM=SUM+COEF(J)*X(JJ)
SUM=SUM-DET                                       00004690
40 SIG=SIG+SUM*SUM
SIG=DSQRT(SIG/M)                                 00004700
RETURN
END                                              00004710
00004720
00004730
00004740
00004750
00004760
00004770
```

G LEVEL 21	SIGNF	DATE = 78102	10/47/48
SUBROUTINE SIGNF(COFB,NCOFS,DET,XPTS,YPTS,ZPTS,NPTS,IFIT)		00004780	
REAL*8 VNORM(3),XVECT(3),XMAG,VMAG,DSQRT		00004781	
REAL*8 COFB(I),DET,XPTS(1),YPTS(1),ZPTS(1)		00004790	
DIMENSION IFIT(1)		00004791	
COSANG=0.		00004792	
DO 20 L=1,NPTS		00004793	
XMAG=DSQRT(XPTS(L)+YPTS(L)+ZPTS(L)+ZPTS(L)+ZPTS(L))		00004794	
IF(XMAG.LT.1.0D-10) GO TO 20		00004795	
XVECT(1)=XPTS(L)/XMAG		00004796	
XVECT(2)=YPTS(L)/XMAG		00004797	
XVECT(3)=ZPTS(L)/XMAG		00004798	
DO 1 I=1,3		00004799	
1 VNORM(I)=0.		00004800	
I=0		00004801	
IF(IFIT(1).EQ.0) GO TO 2		00004802	
I=I+1		00004803	
VNORM(1)=VNORM(1)+COFB(I)		00004804	
IF(I.EQ.NCOFS) GO TO 10		00004805	
2 IF(IFIT(2).EQ.0) GO TO 3		00004806	
I=I+1		00004807	
VNORM(2)=VNORM(2)+COFB(I)		00004808	
IF(I.EQ.NCOFS) GO TO 10		00004809	
3 IF(IFIT(3).EQ.0) GO TO 4		00004810	
I=I+1		00004811	
VNORM(3)=VNORM(3)+COFB(I)		00004812	
IF(I.EQ.NCOFS) GO TO 10		00004813	
4 IF(IFIT(4).EQ.0) GO TO 5		00004814	
I=I+1		00004815	
VNORM(1)=VNORM(1)+COFB(I)*YPTS(L)		00004816	
VNORM(2)=VNORM(2)+COFB(I)*XPTS(L)		00004817	
IF(I.EQ.NCOFS) GO TO 10		00004818	
5 IF(IFIT(5).EQ.0) GO TO 6		00004819	
I=I+1		00004820	
VNORM(1)=VNORM(1)+COFB(I)*ZPTS(L)		00004821	
VNORM(3)=VNORM(3)+COFB(I)*XPTS(L)		00004822	
IF(I.EQ.NCOFS) GO TO 10		00004823	
6 IF(IFIT(6).EQ.0) GO TO 7		00004824	
I=I+1		00004825	
VNORM(2)=VNORM(2)+COFB(I)*ZPTS(L)		00004826	
VNORM(3)=VNORM(3)+COFB(I)*YPTS(L)		00004827	
IF(I.EQ.NCOFS) GO TO 10		00004828	
7 IF(IFIT(7).EQ.0) GO TO 8		00004829	
I=I+1		00004830	
VNORM(1)=VNORM(1)+2.*COFB(I)*XPTS(L)		00004831	
IF(I.EQ.NCOFS) GO TO 10		00004832	
8 IF(IFIT(8).EQ.0) GO TO 9		00004833	
I=I+1		00004834	
VNORM(2)=VNORM(2)+2.*COFB(I)*YPTS(L)		00004835	
IF(I.EQ.NCOFS) GO TO 10		00004836	
9 IF(IFIT(9).EQ.0) GO TO 10		00004837	
I=I+1		00004838	
VNORM(3)=VNORM(3)+2.*COFB(I)*ZPTS(L)		00004839	

G LEVEL	21	SIGNF	DATE = 78102	10/47/48
10	VMAG=DSQRT(VNORM(1)*VNORM(1)+VNORM(2)*VNORM(2)+VNORM(3)*		00004840	
1	VNORM(3))		00004841	
	IF(VMAG.EQ.0) GO TO 1000		00004842	
DO 11 I=1,3			00004843	
11	VNORM(I)=VNORM(I)/VMAG		00004844	
DO 12 I=1,3			00004845	
12	COSANG=COSANG+VNORM(I)*XVECT(I)		00004846	
20	CONTINUE		00004847	
	IF(COSANG.EQ.0) GO TO 2000		00004848	
	IF(COSANG.GT.0) RETURN		00004849	
DO 28 J=1,NCOFS			00004850	
28	COFB(J)=-COFB(J)		00004851	
	DET=-DET		00004852	
	RETURN		00004853	
1000	WRITE(6,200)		00004854	
200	FORMAT(IH1,*MAGNITUDE OF NORMAL VECTOR IS ZERO IN SIGNF*)		00004855	
	STOP		00004856	
2000	WRITE(6,201)		00004857	
201	FORMAT(IH0,*WARNING TEST IN SIGNF SHOWS VECTORS PERPENDICULAR*)		00004858	
	RETURN		00004859	
	END		00004860	

G LEVEL 21	EQNSOL	DATE = 78102	10/47/48
SUBROUTINE EQNSOL(A,B,N,X,DET,NHOMO)		00005000	
IMPLICIT REAL*8 (A-H,O-S,U-Z)		00005010	
REAL*8 TOL		00005020	
DIMENSION A(9,9),B(1),X(1)		00005030	
COMMON/TOLRS/TOLEQN,TOLSNT,TOLQPT,TOLINT,TOLCK1,TOLCK2,		00005040	
1 TOLCK4,TDELT,TOLCIR,TANGS2,TOLPLT		00005050	
TOL=TOLEQN		00005060	
NRANK=N		00005070	
DO 10 I=1,N		00005080	
10 X(I)=0.0		00005090	
I=0		00005100	
ICOL=0		00005110	
12 I=I+1		00005120	
ICOL=ICOL+1		00005130	
IP1=I+1		00005140	
14 AMAXI=A(I,ICOL)		00005150	
L=I :		00005160	
IF(IP1.GT.N) GO TO 150		00005170	
DO 15 K=IP1,N		00005180	
IF(DABS(A(K,ICOL)).GT.DABS(AMAXI)) L=K		00005190	
15 AMAXI=A(L,ICOL)		00005200	
150 IF(DABS(AMAXI).GT.TOL) GO TO 16		00005210	
X(ICOL)=1.0		00005220	
NRANK=NRANK-1		00005230	
IF(ICOL.EQ.N) GO TO 45		00005240	
ICOL=ICOL+1		00005250	
GO TO 14		00005260	
16 IF(L.EQ.I) GO TO 25		00005270	
DO 20 J=1,N		00005280	
SWAP=A(I,J)		00005290	
A(I,J)=A(L,J)		00005300	
20 A(L,J)=SWAP		00005310	
SWAP=B(I)		00005320	
B(I)=B(L)		00005330	
B(L)=SWAP		00005340	
25 XDIV=A(J,ICOL)		00005350	
DO 30 J=ICOL,N		00005360	
30 A(I,J)=A(I,J)/XDIV		00005370	
B(I)=B(I)/XDIV		00005380	
IF(IP1.GT.N) GO TO 42		00005390	
DO 40 K=IP1,N		00005400	
XMUL=A(K,ICOL)		00005410	
DO 35 J=ICOL,N		00005420	
35 A(K,J)=A(K,J)-XMUL*A(I,J)		00005430	
40 B(K)=B(K)-XMUL*B(I)		00005440	
IF(ICOL.LT.N) GO TO 12		00005450	
42 IF(NRANK.EQ.N.OR.NHOMO.EQ.0) X(ICOL)=B(I)/A(I,ICOL)		00005460	
45 IJUMP=ICOL-I		00005470	
DO 50 I=2,N		00005480	
K=N-I+1		00005490	
IF(X(K).EQ.0.0) GO TO 47		00005500	
IJUMP=IJUMP-1		00005510	

G LEVEL	21	EQNSOL	DATE = 78102	10/47/48
	GO TO 50			00005520
47	IF(NRANK.EQ.N.OR.NHOMO.EQ.0) X(K)=B(K)			00005530
	KP1=K+1			00005540
	DO 48 J=KP1,N			00005550
48	X(K)=X(K)-A(K-IJUMP,J)*X(J)			00005560
50	CONTINUE			00005570
	DET=1.0			00005580
	IF(NRANK.LT.N) DET=0.0			00005590
	RETURN			00005600
	END			00005610

G LEVEL 21	CFITS	DATE = 78102	10/47/48
SUBROUTINE CFITS(DATAC,DATAS,ISTART,NPTS,COEFSB,ICOF,SIG,XCURF,		00005620	
1 FCURF)		00005630	
DIMENSION DATAC(1),DATAS(1),COEFSB(1),IFIT(9)		00005640	
DIMENSION SIGMA(8),DETERM(8)		00005650	
REAL*8 XCURF(1),FCURF(1),COFB(9),SIG,DET		00005660	
REAL*8 XPTS(30),YPTS(30),ZPTS(30)		00005661	
IDEFLT=DATAC(1)+0.5		00005670	
IPASS=0		00005680	
CALL NORM(DATAS,ISTART,NPTS,TNORM)		00005690	
L=ISTART-21		00005700	
DO 5 I=1,9		00005710	
5 IFIT(I)=DATAS(L+I)+0.5		00005720	
IF(IDEFLT.EQ.0) GO TO 8		00005721	
DO 6 I=1,3		00005722	
6 IFIT(I)=1		00005723	
8 CONTINUE		00005724	
NCOEFS=0		00005730	
DO 10 I=1,9		00005740	
10 IF(IFIT(I).EQ.1) NCOEFS=NCOEFS+1		00005750	
IF(NPTS.LT.NCOEFS) GO TO 50		00005760	
15 CONTINUE		00005770	
IPASS=IPASS+1		00005780	
DO 35 L=1,NPTS		00005790	
LJ=(L-1)+NCOEFS		00005800	
IJ=(L-1)*3+ISTART		00005810	
XV=DATAS(IJ)/TNORM		00005820	
YY=DATAS(IJ+1)/TNORM		00005830	
ZV=DATAS(IJ+2)/TNORM		00005840	
XPTS(L)=XV		00005841	
YPTS(L)=YY		00005842	
ZPTS(L)=ZV		00005843	
I=0		00005860	
IF(IFIT(1).EQ.0) GO TO 310		00005870	
I=I+1		00005880	
XCURF(LJ+I)=XV		00005890	
310 IF(IFIT(2).EQ.0) GO TO 311		00005900	
I=I+1		00005910	
XCURF(LJ+I)=YY		00005920	
311 IF(IFIT(3).EQ.0) GO TO 312		00005930	
I=I+1		00005940	
XCURF(LJ+I)=ZV		00005950	
312 IF(IFIT(4).EQ.0) GO TO 313		00005960	
I=I+1		00005970	
XCURF(LJ+I)=XV+YY		00005980	
313 IF(IFIT(5).EQ.0) GO TO 314		00005990	
I=I+1		00006000	
XCURF(LJ+I)=XV+ZV		00006010	
314 IF(IFIT(6).EQ.0) GO TO 315		00006020	
I=I+1		00006030	
XCURF(LJ+I)=YY+ZV		00006040	
315 IF(IFIT(7).EQ.0) GO TO 316		00006050	
I=I+1		00006060	

6 LEVEL	21	CFITS	DATE = 78102	10/47/48
		XCURF(LJ+I)=XV*XV		00006070
316		IF(IFIT(8).EQ.0) GO TO 317		00006080
	I=I+1			00006090
		XCURF(LJ+I)=YV*YV		00006100
317		IF(IFIT(9).EQ.0) GO TO 35		00006110
	I=I+1			00006120
		XCURF(LJ+I)=ZV*ZV		00006130
35		FCURF(L)=1.		00006140
		CALL CURFIT(FCURF,XCURF,NCOEFS,NPTS,CQFB,DET,SIG,1)		00006160
		CALL SIGNF(CQFB,NCOEFS,DET,XPTS,YPTS,ZPTS,NPTS,IFIT)		00006170
		IF(IDEFLT.EQ.0) GO TO 350		00006180
		IF(IPASS.EQ.9) GO TO 350		00006190
		SIGMA(IPASS)=SIG		00006200
		DETERM(IPASS)=DET		00006210
		GO TO (318,319,320,321,322,323,324,325), IPASS		00006220
318		IF(NPTS.LE.3) GO TO 350		00006230
	NCOEFS=4			00006240
	IFIT(7)=1			00006250
		GO TO 15		00006260
319		IFIT(7)=0		00006270
	IFIT(8)=1			00006280
		GO TO 15		00006290
320		IFIT(8)=0		00006300
	IFIT(9)=1			00006310
		GO TO 15		00006320
321		IF(NPTS.LE.4) GO TO 325		00006330
	NCOEFS=5			00006340
	IFIT(7)=1			00006350
	IFIT(8)=1			00006360
	IFIT(9)=0			00006370
		GO TO 15		00006380
322		IFIT(8)=0		00006390
	IFIT(9)=1			00006400
		GO TO 15		00006410
323		IFIT(7)=0		00006420
	IFIT(8)=1			00006430
		GO TO 15		00006440
324		IF(NPTS.LE.5) GO TO 325		00006450
	NCOEFS=6			00006460
	IFIT(7)=1			00006470
	GO TO 15			00006480
325		IMATCH=0		00006490
3250		SIG=SIGMA(1)		00006500
	L=1			00006510
	NCOEFS=8			00006520
	IF(NPTS.EQ.4) NCOEFS=4			00006530
	IF(NPTS.EQ.5) NCOEFS=7			00006540
	DO 326 I=2,NCOEFS			00006550
	IF(ABS(DETERM(I)).LT.0.5.AND.IMATCH.EQ.0) GO TO 326			00006560
	IF(SIGMA(I).LT.SIG) L=I			00006570
	IF(L.EQ.I) SIG=SIGMA(I)			00006580
326		CONTINUE		00006590

G LEVEL	21	CFITS	DATE = 78102	10/47/48
		IF(IMATCH.EQ.0.AND.SIG.LT.1.0D-08) GO TO 327	00006600	
		IF(IMATCH.EQ.1.AND.SIG.LT.1.0D-06) GO TO 327	00006610	
		IF(IMATCH.EQ.1) GO TO 60	00006620	
		IMATCH=1	00006630	
		GO TO 3250	00006640	
327	IFIT(7)=0	IFIT(8)=0	00006650	
		IFIT(9)=0	00006660	
		NCOEFS=3	00006670	
		IF(L.GT.1) NCOEFS=4	00006680	
		IF(L.GT.4) NCOEFS=5	00006690	
		IF(L.EQ.8) NCOEFS=6	00006700	
		IPASS=8	00006710	
		GO TO (15,328,329,330+331,332,333,334), L	00006720	
328	IFIT(7)=1	IFIT(8)=0	00006730	
		GO TO 15	00006740	
329	IFIT(8)=1	GO TO 15	00006750	
330	IFIT(9)=1	GO TO 15	00006760	
331	IFIT(7)=1	IFIT(8)=1	00006770	
		GO TO 15	00006780	
332	IFIT(7)=1	IFIT(8)=1	00006790	
		GO TO 15	00006800	
		IFIT(9)=1	00006810	
		GO TO 15	00006820	
333	IFIT(8)=1	IFIT(9)=1	00006830	
		GO TO 15	00006840	
334	IFIT(7)=1	IFIT(8)=1	00006850	
		GO TO 15	00006860	
		IFIT(9)=1	00006870	
		GO TO 15	00006880	
335	IFIT(7)=1	IFIT(8)=1	00006890	
		GO TO 15	00006900	
		IFIT(9)=1	00006910	
		GO TO 15	00006920	
350	CONTINUE	L=(ICOF-1)*10	00006930	
		J=0	00006940	
		DO 40 I=1,10	00006950	
		COEFSB(L+I)=0.	00006960	
		IF(I.EQ.10) GO TO 36	00006970	
		IF(IFIT(I).EQ.0) GO TO 40	00006980	
		J=J+1	00006990	
36	COEFSB(L+I)=DATAS(ISTART+I-12)	IF(IFIT(I).EQ.0) GO TO 40	00007010	
		IF(COEFSB(L+I).NE.0.) GO TO 40	00007020	
		IF(I.EQ.10) GO TO 38	00007030	
		COEFSB(L+I)=COFB(J)/TNORM	00007C40	
		IF(I.LE.3) GO TO 40	00007050	
		COEFSB(L+I)=COEFSB(L+I)/TNORM	00007060	
		GO TO 40	00007070	
38	COEFSB(L+I)=-DET	00007080		
40	CONTINUE	00007090		
	RETURN	00007100		
50	WRITE(6,100) ICOF	00007110		

G LEVEL 21

CFITS

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100 FORMAT(1H1,*REFLECTIVE SURFACE NO. *,I3,* HAS NOT BEEN DEFINED BY 00007120
      1ENOUGH POINTS*)                                              . 00007130
      STOP                                                       00007140
  60 WRITE(6,110) I05F
110 FORMAT(1H1,*BOUNDARY POINTS FOR REFLECTIVE SURFACE NO. *,I3,
      1 * DO NOT FORM A CURVE-FIT WITHIN ERROR TOLERANCE*/        00007160
      2 1H0,*PROGRAM WILL CONTINUE WITH BEST FIT POSSIBLE*)       00007170
      GO TO 327                                                 00007180
      END                                                       00007190
                                                               00007200
```

G LEVEL 21	OUTPT2	DATE = 78102	10/47/48
	SUBROUTINE OUTPT2(COEFB,SIG,NSETS,TITLES,LOOP)	00007210	
	DIMENSION COEFB(1),TITLES(1),SIG(1)	00007220	
	WRITE(6,102) (TITLES(MM),MM=1,80)	00007230	
102	FORMAT(1H1,80A1)	00007240	
	WRITE(6,103)	00007250	
103	FORMAT(1H0,46X,'*** RESULTS OF CURVE-FIT ANALYSIS ***')	00007260	
	GO TO (10,50), LOOP	00007270	
10	CONTINUE	00007280	
	WRITE(6,110)	00007290	
110	FORMAT(1H0,2X,'**FENCE DATA*')	00007300	
	LINES=5	00007310	
	DO 20 I=1,NSETS	00007320	
	LL=(I-1)*4	00007330	
	IF((LINES+6).LE.50) GO TO 15	00007340	
	WRITE(6,102) (TITLES(MM),MM=1,80)	00007350	
	WRITE(6,103)	00007360	
	WRITE(6,110)	00007370	
	LINES=5	00007380	
15	WRITE(6,112) I	00007390	
112	FORMAT(1H0,5X,'**FENCE NO.',I3)	00007400	
	LINES=LINES+2	00007410	
	WRITE(6,121) (COEFB(LL+L),L=1,4)	00007420	
121	FORMAT(1H0,9X,'VALUES FOR COEFFICIENTS OF SURFACE*/1H ,1IX,	00007430	
	1 *(1) X COEF *,E12.4,5X,* (2) Y COEF *,E12.4,5X,* (3) Z COEF *00007440		
	2,E12.4,5X,* (4) CONSTANT *,E12.4)	00007450	
	WRITE(6,123) SIG(I)	00007460	
123	FORMAT(1H0,9X,'ROOT MEAN SQUARE ERROR OF FIT=*,E12.4)	00007470	
	LINES=LINES+5	00007480	
20	CONTINUE	00007490	
	RETURN	00007500	
50	CONTINUE	00007510	
	WRITE(6,117)	00007520	
117	FORMAT(1H0,2X,'**REFLECTIVE SURFACE DATA*')	00007530	
	LINES=5	00007540	
	LT=0	00007550	
	DO 60 I=1,NSETS	00007560	
	LT=LT+80	00007570	
	LL=(I-1)*10	00007580	
	IF((LINES+9).LE.50) GO TO 55	00007590	
	WRITE(6,102) (TITLES(MM),MM=1,80)	00007600	
	WRITE(6,103)	00007610	
	WRITE(6,117)	00007620	
	LINES=5	00007630	
55	WRITE(6,118) I,(TITLES(LT+MM),MM=1,80)	00007640	
118	FORMAT(1H0,5X,'*REFLECTIVE SURFACE NO.',I3,5X,'IDENTIFICATION: ',00007650		
	1 80A1)	00007660	
	LINES=LINES+2	00007670	
	WRITE(6,122) (COEFB(LL+L),L=1,10)	00007680	
122	FORMAT(1H0,9X,'VALUES FOR COEFFICIENTS OF SURFACE*/1H ,1IX,* (1) X 00007690		
	1 COEF *,E12.4,5X,* (2) Y COEF *,E12.4,5X,* (3) Z COEF *,E12.4/ 00007700		
	2 1H ,1IX,* (4) XY COEF *,E12.4,5X,* (5) XZ COEF *,E12.4,5X, 00007710		
	3 *(6) YZ COEF *,E12.4/1H ,1IX,* (7) XSQ COEF *,E12.4,5X, 00007720		

G LEVEL 21	OUTPT2	DATE = 78102	10/47/48
4 *(8) YSQ COEF *,E12.4,5X,*(9) ZSQ COEF *,E12.4/1H +10X*			00007730
5 *(10) CONSTANT *,E12.4)			00007740
WRITE(6,123) SIG(I)			00007750
LINES=LINES+8			00007760
60 CONTINUE			00007770
RETURN			00007780
END			00007790

G LEVEL 21

GLINT

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SUBROUTINE GLINT(DATAAC, DATAAS, MS, COEFSB, GAMMA, ALPHA2, BETA2, INTRFL, 00007800
1 ISHADW, INTERF, DWORKS, DATAF, MF) 00007810
DIMENSION ALPHA2(1), BETA2(1), INTRFL(1), ISHADW(1), INTERF(1) 00007820
DIMENSION DATAAC(1), DATAAS(1), MS(1), COEFSB(1) 00007830
DIMENSION DWORKS(1), DATAF(1), MF(1) 00007840
DIMENSION SUN(3), DIRCOS(3,3), XB(3), SUNB(3), VNORMB(3), X(3), 00007850
1 RFLTNB(3), RFLTN(3) 00007860
DATA ONE,ZERO,RAD/1.0E00,0.0E00,57.29578E00/ 00007870
SUN(1)=-COS(GAMMA/RAD) 00007880
SUN(2)=0. 00007890
SUN(3)=-SIN(GAMMA/RAD) 00007900
DQ2=DATAAC(15)*DATAAC(15)*144. 00007910
THETA=DATAAC(4) 00007920
PHI=DATAAC(5) 00007930
SNTH=SIN(THETA/RAD) 00007940
CSTH=COS(THETA/RAD) 00007950
SNPH=SIN(PHI/RAD) 00007960
CSPH=COS(PHI/RAD) 00007970
SPST=SNPH*SNTH 00007980
SPCT=SNPH*CSTH 00007990
CPST=CSPH*SNTH 00008000
CPCT=CSPH*CSTH 00008010
DIRCOS(1,3)=-SNTH 00008020
DIRCOS(2,3)=-SPCT 00008030
DIRCOS(3,3)=CPCT 00008040
DPSI=DATAAC(11) 00008050
NPSI=DATAAC(9)+0.5 00008060
NFENCE=DATAAC(22)+0.5 00008070
NPANEL=DATAAC(23)+0.5 00008080
MAS=0 00008090
DO 100 I=1,NPANEL 00008100
N10=(I-1)*10 00008110
KI=MS(I) 00008120
NPTS=DATAS(KI)+0.5 00008130
PSIKP=DATAAC(10)-DPSI 00008140
DO 60 J=1,NPSI 00008150
PSIKP=PSIKP+DPSI 00008160
PSI=PSIKP 00008170
IF(PSIKP.LT.-179.) PSI=-179. 00008180
IF(PSIKP.GT.179.) PSI=179. 00008190
IF(PSIKP.GT.-91.AND.PSIKP.LT.-89.) PSI=-89. 00008200
IF(PSIKP.LT.91.AND.PSIKP.GT.89.) PSI=89. 00008210
SNPS=SIN(PSI/RAD) 00008220
CSPS=COS(PSI/RAD) 00008230
DIRCOS(1,1)=-CSTH*CSPS 00008240
DIRCOS(1,2)=-CSTH*SNPS 00008250
DIRCOS(2,1)=SPST*CSPS-SNPS*CSPH 00008260
DIRCOS(2,2)=SPST*SNPS+CSPS*CSPH 00008270
DIRCOS(3,1)=-CPST*CSPS-SNPS*SNPH 00008280
DIRCOS(3,2)=-CPST*SNPS+SNPH*CSPS 00008290
DO 20 M=1,3 00008300
SUNB(M)=0. 00008310

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G LEVEL	21	GLINT	DATE = 78102	10/47/48
	DO 20 K=1,3			00008320
20	SUNB(M)=SUNB(M)+DIRCOS(M,K)*SUN(K)			00008330
	DO 75 L=1,NPTS			00008340
	MAS=MAS+1			00008350
	INTRFL(MAS)=0			00008360
	ISHADW(MAS)=0			00008370
	INTERF(MAS)=0			00008380
	N3=(L-1)*3+KI			00008390
	DO 10 M=1,3			00008400
10	XB(M)=DATAS(N3+M)			00008410
	IF(NFENCE.GT.0) CALL SHADOW(DATAF, MF, NFENCE, XB, SUNB, ISHADW,			00008420
	1 MAS,DWORKS)			00008430
	VNORMB(1)=COEFSB(N10+1)+COEFSB(N10+4)*XB(2)+COEFSB(N10+5)*XB(3)+			00008440
	1 2.*COEFSB(N10+7)*XB(1)			00008450
	VNORMB(2)=COEFSB(N10+2)+COEFSB(N10+4)*XB(1)+COEFSB(N10+6)*XB(3)+			00008460
	1 2.*COEFSB(N10+8)*XB(2)			00008470
	VNORMB(3)=COEFSB(N10+3)+COEFSB(N10+5)*XB(1)+COEFSB(N10+6)*XB(2)+			00008480
	1 2.*COEFSB(N10+9)*XB(3)			00008490
	ASN=0.			00008500
	ABSNQ=0.			00008510
	DO 30 M=1,3			00008520
	ASN=ASN+SUNB(M)+VNORMB(M)			00008530
30	ABNSQ=ABNSQ+VNORMB(M)**2			00008540
	ASN=ASN/SQRT(ABNSQ)			00008550
	IF(ASN.GT.0.001745) INTRFL(MAS)=1			00008560
	DO 40 M=1,3			00008570
40	RFLTNB(M)=SUNB(M)-2.*ASN*VNORMB(M)/SQRT(ABNSQ)			00008580
	IF(NFENCE.GT.0) CALL INTFER(DATAF, MF, NFENCE, XB, RFLTNB, INTERF,			00008590
	1 MAS,DWORKS)			00008600
	DO 50 M=1,3			00008610
	RFLTN(M)=0.			00008620
	X(M)=0.			00008630
	DO 50 K=1,3			00008640
	RFLTN(M)=RFLTN(M)+DIRCOS(K,M)*RFLTNB(K)			00008650
50	X(M)=X(M)+DIRCOS(K,M)*XB(K)			00008660
	CALL QUADPT(ONE, ONE, ZERO, ZERO, ZERO, ZERO, ZERO, ZERO, ZERO, DQ2,			00008670
	1 RFLTN(1), RFLTN(2), RFLTN(3), X(1), X(2), X(3), XI1, YI1, ZI1, XI2,			00008680
	2 YI2, ZI2, INTC)			00008690
	IF(INTC.EQ.1) GO TO 55			00008700
	CALL SORNOT(RFLTN(1), RFLTN(2), RFLTN(3), X(1), X(2), X(3), XI1+YI1+ZI1,			00008710
	1 ISHAD)			00008720
	IF(ISSHAD.EQ.1) GO TO 55			00008730
	XI1=XI2			00008740
	YI1=YI2			00008750
	ZI1=ZI2			00008760
55	CONTINUE			00008770
	ALPHA2(MAS)=RAD*ATAN2(ZI1,SQRT(XI1*XII+YI1*YII))			00008780
	BETA2(MAS)=RAD*ATAN2(YI1,XI1)			00008790
75	CONTINUE			00008800
60	CONTINUE			00008810
100	CONTINUE			00008820
	RETURN			00008830

G LEVEL 21

GLINT

DATE = 78102

10/47/48

END

00008840

G LEVEL	21	SHADOW	DATE = 78102	10/47/48
		SUBROUTINE SHADOW (DATAF,MF,NFENCE,XBS,SUNB,ISHADW,MAS,DWORKS)	00008850	
		DIMENSION DATAF(1),MF(1),XBS(1),SUNB(1),ISHADW(1),DWORKS(1),XBF(3)	00008860	
		APL=SUNB(1)	00008870	
		BPL=SUNB(2)	00008880	
		CPL=SUNB(3)	00008890	
		DPL=APL+XBS(1)-BPL+XBS(2)-CPL+XBS(3)	00008900	
		THETA=0.	00008910	
		IF(APL.NE.0.0.OR.CPL.NE.0.0) THETA=ATAN2(APL+CPL)	00008920	
		ALPHA=ATAN2(BPL,SQRT(APL+APL+CPL+CPL))	00008930	
		SNTH=SIN(THETA)	00008940	
		CSTH=COS(THETA)	00008950	
		SNAL=SIN(ALPHA)	00008960	
		CSAL=COS(ALPHA)	00008970	
		SAST=SNAL*SNTH	00008980	
		SACT=SNAL*CSTH	00008990	
		CAST=CSAL*SNTH	00009000	
		CACT=CSAL*CSTH	00009010	
		DO 50 I=1,NFENCE	00009020	
		KF=MF(I)	00009030	
		NPTSF=DATAF(KF)+0.5	00009040	
		DO 30 J=1,NPTSF	00009050	
		LI=(J-1)*3	00009060	
		LK=(J-1)*2	00009070	
		DO 10 K=1,3	00009080	
10		XBF(K)=DATAF(LI+K+1)	00009090	
		CALL PLANPT(APL,BPL,CPL,DPL,SUNB(1),SUNB(2)+SUNB(3),DISC,XBF(1),	00009100	
1		1 XBF(2),XBF(3),XINTP,YINTP,ZINTP)	00009110	
		CALL SORNOT(SUNB(1),SUNB(2),SUNB(3),XBF(1),XEF(2),XBF(3),XINTP,	00009120	
1		1 YINTP,ZINTP,ISH)	00009130	
		IF(ISH.EQ.0) GO TO 50	00009140	
20		DWORKS(LK+1)=XINTP*CSTH-ZINTP*SNTH	00009150	
		DWORKS(LK+2)=-XINTP*SAST+YINTP*CSAL-ZINTP*SACT	00009160	
30		CONTINUE	00009170	
		XC=XBS(1)*CSTH-XPS(3)*SNTH	00009180	
		YC=-XBS(1)*SAST+XBS(2)*CSAL-XBS(3)*SACT	00009190	
		CALL CIRCLE(DWORKS,NPTSF,XC,YC,INCR)	00009200	
		IF(INCR.EQ.0) GO TO 50	00009210	
		ISHADW(MAS)=1	00009220	
		GO TO 60	00009230	
50		CONTINUE	00009240	
60		RETURN	00009250	
		END	00009260	

G LEVEL	21	PLANPT	DATE = 78102	10/47/48
		SUBROUTINE PLANPT(A,B,C,D,SL,SM,SN,DISC,X1,Y1,Z1,X,Y,Z)		00009270
		COMMON/TOLRS/TOLEQN,TOLSNT,TOLQPT,TOLINT,TOLCK1,TOLCK2,		00009280
		1 TOLCK4,DELT,TOLCIR,ANGST2,TOL		00009290
		IF(ABS(SL).GE.ABS(SN).AND.ABS(SL).GE.ABS(SM)) DISC=A+B*SM/SL+		00009300
		1 C*SN/SL		00009310
		IF(ABS(SL).LT.ABS(SN).AND.ABS(SM).LE.ABS(SN)) DISC=A*SL/SN+		00009320
		1 B*SM/SM+C		00009330
		IF(ABS(SL).LT.ABS(SM).AND.ABS(SN).LT.ABS(SM)) DISC=A*SL/SM+		00009340
		1 B+C*SN/SM		00009350
		IF(ABS(DISC).LT.TOL) RETURN		00009360
		IF(ABS(SL).LT.ABS(SN).AND.ABS(SM).LE.ABS(SN)) GO TO 10		00009370
		IF(ABS(SL).LT.ABS(SM).AND.ABS(SN).LT.ABS(SM)) GO TO 20		00009380
		X=(-D-B*Y1+B*X1*SM/SL-C*Z1+C*X1*SN/SL)/DISC		00009390
		Y=Y1+SM*(X-X1)/SL		00009400
		Z=Z1+SN*(X-X1)/SL		00009410
		RETURN		00009420
10		Z=(-D-A*X1+A*Z1*SL/SN-B*Y1+B*Z1*SM/SM)/DISC		00009430
		X=X1+SL*(Z-Z1)/SN		00009440
		Y=Y1+SM*(Z-Z1)/SN		00009450
		RETURN		00009460
20		Y=(-D-A*X1+A*SL+Y1/SM-C*Z1+C*SN*Y1/SM)/(A*SL/SM+B+C*SN/SM)		00009470
		X=X1+SL*(Y-Y1)/SM		00009480
		Z=Z1+SN*(Y-Y1)/SM		00009490
		RETURN		00009500
		END		00009510

G LEVEL 21	SORNOT	DATE = 78102	10/47/48
SUBROUTINE SORNOT(SL,SM,SN,X1,Y1,Z1,X,Y,Z,ISHAD)		00009520	
COMMON/TOLRS/TOLEQN,TOL,TOLQPT,TOLINT,TOLCK1,TOLCK2,		00009530	
1 TOLCK4,DELT,TOLCIR,ANGST2,TOLPLT		00009540	
ISHAD=2		00009550	
SL1=X-X1		00009560	
SM1=Y-Y1		00009570	
SN1=Z-Z1		00009580	
IF((ABS(SL1)+ABS(SM1)+ABS(SN1)).LT.TOL) RETURN		00009590	
DIR=0.0		00009600	
IF(ABS(SL).GT.TOL) DIR=SL1/SL		00009610	
IF(ABS(SM).GT.TOL) DIR=DIR+SM1/SM		00009620	
IF(ABS(SN).GT.TOL) DIR=DIR+SN1/SN		00009630	
ISHAD=1		00009640	
IF(DIR.LT.0.0) ISHAD=0		00009650	
RETURN		00009660	
END		00009670	

G LEVEL	21	INTFER	DATE = 78102	10/47/48
		SUBROUTINE INTFER(DATAF,MF,NFENCE,XBS,RFLTNB,INTERF,MAS,DWORKS)	00009680	
		DIMENSION DATAF(1),MF(1),XBS(1),RFLTNB(1)+INTERF(1)+DWORKS(1)+	00009690	
1	XBF(3)		00009700	
		APL=-RFLTNB(1)	00009710	
		BPL=-RFLTNB(2)	00009720	
		CPL=-RFLTNB(3)	00009730	
		DPL=APL*XBS(1)-BPL*XBS(2)-CPL*XBS(3)	00009740	
		THETA=0.	00009750	
		IF(APL.NE.0.0.OR.CPL.NE.0.0) THETA=ATAN2(APL,CPL)	00009760	
		ALPHA=ATAN2(BPL,SQRT(APL*APL+CPL*CPL))	00009770	
		SNTH=SIN(THETA)	00009780	
		CSTH=COS(THETA)	00009790	
		SNAL=SIN(ALPHA)	00009800	
		CSAL=COS(ALPHA)	00009810	
		SAST=SNAL*SNTH	00009820	
		SACT=SNAL*CSTH	00009830	
		CAST=CSAL*SNTH	00009840	
		CACT=CSAL*CSTH	00009850	
	DO 50 I=1,NFENCE		00009860	
	KF=MF(I)		00009870	
	NPTSF=DATAF(KF)+0.5		00009880	
	DO 30 J=1,NPTSF		00009890	
	LI=(J-1)*3		00009900	
	LK=(J-1)*2		00009910	
	DO 10 K=1,3		00009920	
10	XBF(K)=DATAF(LI+K+1)		00009930	
	CALL PLANPT(APL,BPL,CPL,DPL,RFLTNB(1),RFLTNB(2),RFLTNB(3),DISC,		00009940	
1	XBF(1),XBF(2),XBF(3),XINTP,YINTP,ZINTP)		00009950	
	CALL SORNOT(RFLTNB(1),RFLTNB(2),RFLTNB(3),XBF(1),XBF(2),XBF(3),		00009960	
1	XINTP,YINTP,ZINTP,ISH)		00009970	
	IF(ISH.EQ.1) GO TO 50		00009980	
20	DWORKS(LK+1)=XINTP*CSTH-ZINTP*SNTH		00009990	
	DWORKS(LK+2)=-XINTP*SAST+YINTP*CSAL-ZINTP*SACT		00010000	
30	CONTINUE		00010010	
	XC=XBS(1)*CSTH-XBS(3)*SNTH		00010020	
	YC=-XBS(1)*SAST+XBS(2)*CSAL-XRS(3)*SACT		00010030	
	CALL CIRCLE(DWORKS,NPTSF,XC,YC,INCR)		00010040	
	IF(INCR.EQ.0) GO TO 50		00010050	
	INTERF(MAS)=1		00010060	
	GO TO 60		00010070	
50	CONTINUE		00010080	
60	RETURN		00010090	
	END		00010100	

G LEVEL	21	QUADPT	DATE = 78102	10/47/48
		SUBROUTINE QUADPT(A+B+C+D+E+F+G+H+CK+CL,SL,SM,SN,X1,Y1,Z1,XI1,YI1)	00010110	
1	ZI1,XI2,YI2,ZI2,INTC)		00010120	
	COMMON/TOLRS/TOLEGN,TOLSNT,TOL,TOLINT,TOLCK1,TOLCK2,		00010130	
1	TOLCK4,DELT,TOLCIR,ANGST2,TOLPLT		00010140	
	INTC=0		00010150	
	IF(ABS(SL).LT.ABS(SN).AND.ABS(SM).LE.ABS(SN)) GO TO 30		00010160	
	IF(ABS(SL).LT.ABS(SM).AND.ABS(SN).LT.ABS(SM)) GO TO 50		00010170	
	SA=A+B*(SM/SL)**2+C*(SN/SL)**2*D*SM/SL+F*SM*SN/SL**2+E*SN/SL		00010180	
	SB=2.*B*Y1*SM/SL-2.*B*X1*(SM/SL)**2+C*Z1*SN/SL-2.*C*X1*		00010190	
1	(SN/SL)**2*D*Y1-C*X1*SM/SL+E*Z1-E*X1*SN/SL+F*Z1*SM/SL-F*X1*SM*		00010200	
2	SN/SL**2+F*Y1*SN/SL-F*X1*SM*SN/SL**2+G+H*SM/SL+CK*SN/SL		00010210	
	SC=B*Y1*Y1*B*X1*X1*(SM/SL)**2+C*Z1*Z1-2.*B*Y1*X1*SM/SL+C*X1*X1*		00010220	
1	(SN/SL)**2+C*X1*Z1*SN/SL-F*Y1*Z1-F*X1*Y1*SN/SL-F*X1*Z1*SM/SL*		00010230	
2	F*X1*X1*SM*SN/SL**2+H*Y1-H*X1*SM/SL+CK*Z1-CK*X1*SN/SL+CL		00010240	
	ICONTR=1		00010250	
5	DISC=SB*SB-4.*SA*SC		00010260	
	IF(DISC.LT.ABS(SA)/10.) RETURN		00010270	
	IF(DISC.LT.0.0) RETURN		00010280	
	IF(ABS(SA).GE.TOL) GO TO 10		00010290	
	IF(ABS(SB).LT.TOL) RETURN		00010300	
	XI1=-SC/SB		00010310	
	INTC=1		00010320	
	IF(ICONTR.EQ.2) GO TO 60		00010330	
	IF(ICONTR.EQ.3) GO TO 40		00010340	
	GO TO 20		00010350	
10	XI1=-SB/SA/2.-SQRT(DISC)/SA/2.		00010360	
	XI2=-SB/SA/2.-SQRT(DISC)/SA/2.		00010370	
	INTC=2		00010380	
	IF(ICONTR.EQ.2) GO TO 60		00010390	
	IF(ICONTR.EQ.3) GO TO 40		00010400	
20	YI1=Y1+SM*(XI1-X1)/SL		00010410	
	ZI1=Z1+SN*(XI1-X1)/SL		00010420	
	IF(INTC.EQ.1) RETURN		00010430	
	YI2=Y1+SM*(XI2-X1)/SL		00010440	
	ZI2=Z1+SN*(XI2-X1)/SL		00010450	
	RETURN		00010460	
30	SA=C+B*(SM/SN)**2+A*(SL/SN)**2+F*SM/SN+E*SL/SL*SN*D*SM*SL/SN**2		00010470	
	SB=2.*B*Y1*SM/SM-2.*B*Z1*(SM/SN)**2+C*A*X1*SL/SL-2.*A*Z1*		00010480	
1	(SL/SM)**2+F*Y1-F*Z1*SM/SN+E*X1-E*Z1*SL/SL*SN*D*X1*SM/SM-0.*Z1*SM*		00010490	
2	SL/SM**2*D*Y1*SL/SM-D*Z1*SM*SL/SL*SN**2+C*H*SM*SN+G*SL/SL		00010500	
	SC=B*Y1*Y1*B*Z1*Z1*(SM/SM)**2+C*Z1*SM/SM+A*X1*X1+A*Z1*Z1*		00010510	
1	(SL/SM)**2+C*A*X1*Z1*SL/SL*D*Y1*X1-D*Y1*Z1*SL/SL-D*X1*Z1*SM/SM*		00010520	
2	D*Z1*Z1*SM*SL/SL*SN**2+H*Y1-H*Z1*SM/SM+G*X1-G*Z1*SL/SL+CL		00010530	
	ICONTR=3		00010540	
	GO TO 5		00010550	
40	ZI1=XI1		00010560	
	YI1=Y1+SM*(ZI1-Z1)/SN		00010570	
	XI1=X1+SL*(ZI1-Z1)/SN		00010580	
	IF(INTC.EQ.1) RETURN		00010590	
	ZI2=XI2		00010600	
	YI2=Y1+SM*(ZI2-Z1)/SN		00010610	
	XI2=X1+SL*(ZI2-Z1)/SN		00010620	

G LEVEL	21	QUADPT	DATE = 78102	10/47/48
	RETURN			00010630
50	SA=B+A*(SL/SM)**2+C*(SN/SM)**2+D*SL/SM+E*SL*SN/SM**2+F*SN/SM SB=2.*A*X1*SL/SM-2.*A*Y1*(SL/SM)**2+2.*C*Z1*SN/SM-2.*C*Y1*			00010640 00010650
	1 (SN/SM)**2*D*X1-D*Y1*SL/SM+E*X1*SN/SM+E*Z1*SL/SM-2.*E*Y1*SL*SN/ 2 SM**2+F*Z1-F*Y1*SN/SM+G*SL/SM+H*CK*SN/SM SC=A*X1*X1*A*Y1*Y1*(SL/SM)**2-2.*A*Y1*X1*SL/SM+C*Z1*Z1+C*Y1*Y1*			00010660 00010670 00010680
	1 (SN/SM)**2-2.*C*Y1*Z1*SN/SM+E*X1*Z1-E*X1*Y1*SN/SM-E*Y1*Z1*SL/SM+ 2 E*Y1*Y1*SL*SN/SM**2+G*X1-G*Y1*SL/SM+CK*Z1-CK*Y1*SN/SM+CL			00010690 00010700 00010710
	<u>I CONTR=2</u>			
	GO TO 5			00010720
60	YI1=XI1 XI1=X1*SL*(YI1-Y1)/SM ZI1=Z1*SN*(YI1-Y1)/SM IF(INTC.EQ.1) RETURN YI2=XI2 XI2=X1*SL*(YI2-Y1)/SM ZI2=Z1*SN*(YI2-Y1)/SM RETURN END			00010730 00010740 00010750 00010760 00010770 00010780 00010790 00010800 00010810

G LEVEL	21	OUTPT3	DATE = 78102	10/47/48
		SUBROUTINE OUTPT3(DATAAC,DATAS,MS,TITLES,GAMMA,ALPHA2,BETA2,		00010820
		1 INTRFL,ISHADW,INTERF)		00010830
		DIMENSION INTRFL(1),ISHADW(1),INTERF(1)		00010840
		DIMENSION DATAAC(1),DATAS(1),MS(1),ALPHA2(1),BETA2(1),TITLES(1)		00010850
		LT=0		00010860
		MAS=0		00010870
		NPSI=DATAAC(9)+0.5		00010880
		DPSI=DATAAC(11)		00010890
		NPANEL=DATAAC(23)+0.5		00010900
		DO 90 I=1,NPANEL		00010910
		K=MS(I)		00010920
		NPTS=DATAS(K)+0.5		00010930
		WRITE(6,102) (TITLES(MM),MM=1,80)		00010940
		LT=LT+80		00010950
		WRITE(6,101) I,(TITLES(LT+MM),MM=1,80)		00010960
	102	FORMAT(1H1,80A1)		00010970
	101	FORMAT(1H0,51X,'*** SUN GLINT SIGNATURE ***/1H0,5X,'REFLECTIVE SU	00010980	
		IRFACE NO.',I3,5X,'IDENTIFICATION:',80A1)		00010990
		WRITE(6,103) GAMMA,DATAAC(12),DATAAC(4),DATAAC(13),DATAAC(5),		00011000
	1	DATAAC(14),DATAAC(15)		00011010
	103	FORMAT(1H0,5X,'SUN ELEVATION',F12.2,76X,'XROT',F9.2/		00011020
		1 H ,5X,'A/C PITCH ATTITUDE',F7.2,76X,'YROT',F9.2/1H ,5X,'A/C ROLL	00011030	
		2 ATTITUDE',F8.2,76X,'ZROT',F9.2/1H ,106X,'DISTG',F8.2)		00011040
		WRITE(6,104)		00011050
	104	FORMAT(1H0, 5X,'A/C YAW',20X,'BOUNDARY POINTS -- BODY AXES', 7X,		00011060
		X *STATUS FLAGS FOR POINTS',18X,		00011070
	1	*GLINT',1H ,30X,'(STATIONLINE,BUTTLINE,WATERLINE)', 5X,		00011080
		Y *INRFL',4X,*ISHAD',4X,*INTRF',11X,		00011090
	2	*AZIMUTH',3X,*ELEVATION',1H ,36X,*X',9X,*Y',9X,*Z')		00011100
		LINES=13		00011110
		PSI=DATAAC(10)-DPSI		00011120
	DO	60 J=1,NPSI		00011130
		PSI=PSI+DPSI		00011140
		DC 75 L=1,NPTS		00011150
		MAS=MAS+1		00011160
		N3=(L-1)*3+K		00011170
		XB=DATAS(N3+1)+DATAAC(12)		00011180
		YB=DATAS(N3+2)+DATAAC(13)		00011190
		ZB=DATAS(N3+3)+DATAAC(14)		00011200
		WRITE(6,105) PSI,XB,YB,ZB,INTRFL(MAS),ISHADW(MAS),INTERF(MAS),		00011210
	1	BETA2(MAS),ALPHA2(MAS)		00011220
	105	FORMAT(1H0,5X,F7.2,19X,F8.2,2F10.2,10X,I1+8X,I1+8X,I1+13X,F7.2,		00011230
		I 4X,F6.2)		00011240
		LINES=LINES+2		00011250
	75	CONTINUE		00011260
	60	CONTINUE		00011270
	90	CONTINUE		00011280
		RETURN		00011290
		END		00011300

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CANOPY SUN GLINT EVALUATION COMPUTER PROGRAM. (U)
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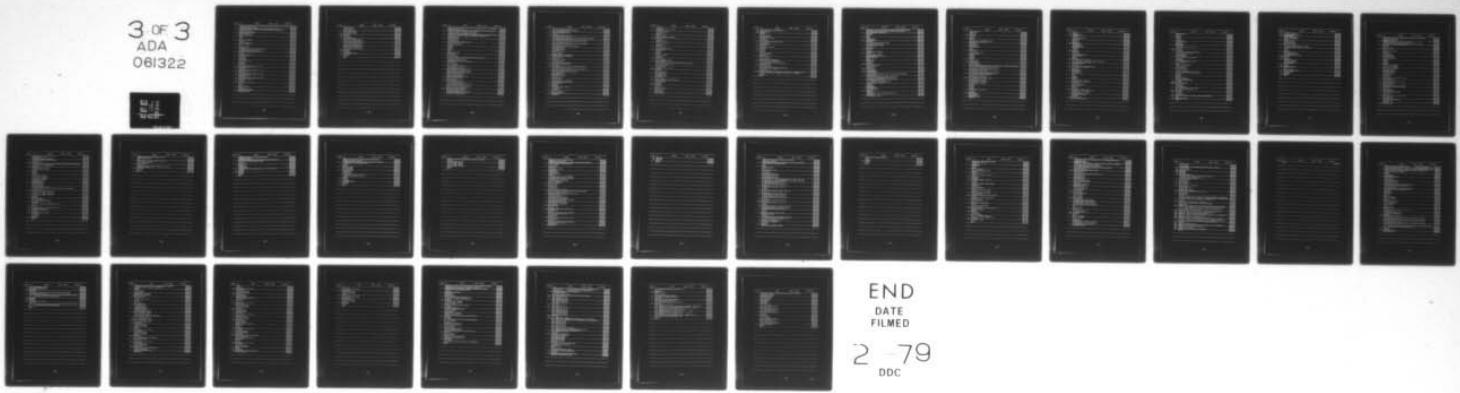
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G LEVEL 21

MINMAX

DATE = 78103

08/54/44

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SUBROUTINE MINMAX(DATAC,DATAS,MS,ALPHA2,BETA2,APLOT,BPLOT,INTRFL,
1 ISHADW,INTERF,IPLTOL) 00011310
DIMENSION DATAC(1),DATAS(1),MS(1),ALPHA2(1),BETA2(1), 00011320
1 APLOT(1),BPLOT(1),INTRFL(1),ISHADW(1),INTERF(1),IPLTOL(1) 00011330
IRFLTN=DATAC(3)+0.5 00011340
NPANEL=DATAC(23)+0.5 00011350
NPSI=DATAC(9)+0.5 00011360
NPSI4=4*(NPSI-1)+1 00011370
IST=1 00011380
MAS=0 00011390
LAS=0 00011400
DO 70 I=1,NPANEL 00011410
KN=MS(I) 00011420
NPTS=DATAS(KN)+0.5 00011430
DO 60 J=1,NPSI 00011440
NP=0 00011450
DO 50 L=1,NPTS 00011460
MAS=MAS+1 00011470
IF(INTRFL(MAS).NE.IRFLTN) GO TO 50 00011480
IF(ISHADW(MAS).EQ.1) GO TO 50 00011490
IF(INTERF(MAS).EQ.1) GO TO 50 00011500
NP=NP+1 00011510
IF(NP.GT.1) GO TO 10 00011520
BLOW=BETA2(MAS) 00011530
KBMIN=MAS 00011540
BHIGH=BLOW 00011550
KBMAX=MAS 00011560
ALOW=ALPHA2(MAS) 00011570
KAMIN=MAS 00011580
AHIGH=ALOW 00011590
KAMAX=MAS 00011600
10 IF(BETA2(MAS).GE.BLOW) GO TO 20 00011610
BLOW=BETA2(MAS) 00011620
KBMIN=MAS 00011630
20 IF(BETA2(MAS).LE.BHIGH) GO TO 30 00011640
BHIGH=BETA2(MAS) 00011650
KBMAX=MAS 00011660
30 IF(ALPHA2(MAS).GE.ALLOW) GO TO 40 00011670
ALLOW=ALPHA2(MAS) 00011680
KAMIN=MAS 00011690
50 CONTINUE 00011700
IF(NP.GT.0) GO TO 55 00011710
AAVG=0. 00011720
BAVG=0. 00011730
DO 52 L=1,NPTS 00011740
M=MAS-L+1 00011750
AAVG=AAVG+BETA2(M) 00011760
52 AAVG=AAVG/NPTS 00011770
AAVG=AAVG/NPTS 00011780
00011790
AAVG=0. 00011800
BAVG=0. 00011810
DO 52 L=1,NPTS 00011820
M=MAS-L+1 00011830
AAVG=AAVG+BETA2(M) 00011840
52 AAVG=AAVG+ALPHA2(M) 00011850
AAVG=AAVG/NPTS 00011860

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G LEVEL 21	MINMAX	DATE = 78103	08/54/44
BAVG=BAVG/NPTS			00011870
DO 54 L=1,4			00011880
IPLTOL(LAS+L)=1			00011890
APLOT(LAS+L)=AAVG			00011900
54 BPLOT(LAS+L)=BAVG			00011910
GO TO 58			00011920
55 CONTINUE			00011930
DO 56 L=1,4			00011940
56 IPLTOL(LAS+L)=0			00011950
APLOT(LAS+1)=AHIGH+0.25			00011960
BPLOT(LAS+1)=BETA2(KAMAX)			00011970
BPLOT(LAS+2)=BLCW-0.25			00011980
APLOT(LAS+2)=ALPHA2(KBMIN)			00011990
APLOT(LAS+3)=ALOW-0.25			00012000
RPLOT(LAS+3)=BETA2(KAMIN)			00012010
BPLOT(LAS+4)=BHIGH+0.25			00012020
APLOT(LAS+4)=ALPHA2(KBMAX)			00012030
58 LAS=LAS+4			00012040
60 CONTINUE			00012050
IST=NPSI4+4			00012060
NPSI4=NPSI4+4+NPSI			00012070
70 CONTINUE			00012100
RETURN			00012110
END			00012120

G LEVEL 21

PROBL

DATE = 78103

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SUBROUTINE PROBL(DATAAC, DATAS, MS, APLOT, BPLOT, XPLOT, YPLOT, DWORKS, 00012130
1 DWKSP2, VPROBL, IPLTOL) 00012140
DIMENSION DATAAC(1), DATAS(1), MS(1), APLOT(1), BPLOT(1), XPLOT(1), 00012150
1 YPLOT(1), DWORKS(1), DWKSP2(1), IPLTOL(1) 00012160
DIMENSION RPS(200), RPSN(200), SPS(200) 00012170
DIMENSION RPSINP(500), NINP(20), RPSOUT(500), NOUT(20), RPSWK1(500). 00012180
1 NWK1(20), RPSWK2(500), NWK2(20), RPSWK3(500), NWK3(20) 00012190
ALOW=DATAAC(16) 00012200
AHIGH=DATAAC(17) 00012210
PETH=DATAAC(19) 00012220
PETH=DATAAC(18) 00012230
TAREA=(AHIGH-ALOW)*(EETH-BETL) 00012240
NPANEL=DATAAC(23)+0.5 00012250
NPSI=DATAAC(9)+0.5 00012260
NCUMI=C 00012270
NSURFI=0 00012280
SWEEP=0. 00012290
LASP=0 00012300
LAS=0 00012310
DO 1A I=1,NPANEL 00012320
NPSI=NPSI 00012330
NPSIS=NPSI+1 00012340
DO 8 J=1,NPSI 00012350
IF(IPLTOL(LASP+1).EQ.0.0,AND,NPSIS.GT.NPSI) NPSI=J 00012360
IF(NPSIS.GT.NPSI) GO TO 8 00012370
IF(NPSIP.LT.NPSI) GO TO 8 00012380
IF(IPLTOL(LASP+1).EQ.1) NPSIP=J-1 00012390
P LASP=LASP+4 00012400
DO 12 J=1,NPSI 00012410
IF(J.LT.NPSIS) GO TO 12 00012420
IFI(J.GT.NPSIP) GO TO 12 00012430
IFI((NPSIP-NPSIS).EQ.0) GO TO 9 00012440
IFI(J.GT.NPSIS) GO TO 10 00012450
BAVG1=PPLOT(LAS+0)+PPLOT(LAS+2) 00012460
BAVG2=PPLOT(LAS+P)+PPLOT(LAS+6) 00012470
XPLOT(1)=BPLOT(LAS+4) 00012480
YPLOT(1)=APLOT(LAS+4) 00012490
IFI(BAVG1.GE.BAVG2) GO TO 10 00012500
9 XPLOT(1)=BPLOT(LAS+2) 00012510
YPLOT(1)=APLOT(LAS+2) 00012520
10 LP=J-NPSIS+2 00012530
XPLOT(LP)=BPLOT(LAS+1) 00012540
YPLOT(LP)=APLOT(LAS+1) 00012550
IFI((NPSIP-NPSIS).EQ.0) GO TO 100 00012560
IFI(J.LT.NPSIP) GO TO 11 00012570
BAVG1=PPLOT(LAS+0)+PPLOT(LAS+2) 00012580
BAVG2=PPLOT(LAS)+BPLOT(LAS-2) 00012590
100 XPLOT(LP+1)=BPLOT(LAS+4) 00012600
YPLOT(LP+1)=APLOT(LAS+4) 00012610
IFI((NPSIP-NPSIS).EQ.0) GO TO 11 00012620
IFI(BAVG1.GE.BAVG2) GO TO 11 00012630
XPLOT(LP+1)=BPLOT(LAS+2) 00012640

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G LEVEL	21	PROBL	DATE = 78103	08/54/44
		YPLOT(LP+1)=APLOT(LAS+2)		00012650
11		LP=2*(NPSIP-NPSIS+1)+3-(J-NPSIS+1)		00012660
		XPLOT(LP)=APLOT(LAS+3)		00012670
		YPLOT(LP)=APLOT(LAS+3)		00012680
12		LAS=LAS+4		00012690
		IF(NPSIS.GT.NPST) GO TO 18		00012700
		NCFS=2*(NPSIP-NPSIS+1)+2		00012710
		CALL SORT2(XPLOT,YPLOT,NCFS,NCFS1,NCFS2,DWORKS,DWKSP2)		00012720
		IF(NCFS2.EQ.0) GO TO 16		00012730
		CALL SORT(YPLOT,XPLOT,NCFS1,BETH,BETL,DWKSP2)		00012740
		IF(NCFS1.EQ.0) GO TO 14		00012750
		CALL SORT(XPLOT,YPLOT,NCFS1,AHIGH,ALOW,DWKSP2)		00012760
		IF(NCFS1.EQ.0) GO TO 14		00012770
		CALL AREA(XPLOT,YPLOT,NCFS1,YINTG)		00012780
		SWEET=SWEET+YINTG		00012790
		NSURFI=NSURFI+1		00012800
		NINP(NSURFI)=NCFS1		00012810
		N2=2*NCFS1		00012820
		DO 13 L=1,NCFS1		00012830
		J2=(L-1)*2+1		00012840
		RPSINP(NCUMI+J2)=XPLOT(L)		00012850
13		RPSINP(NCUMI+J2+1)=YPLOT(L)		00012860
		NCUMI=NCUMI+N2		00012870
14		CONTINUE		00012880
		DO 15 KKK=1,NCFS2		00012890
		LML=(KKK-1)*2		00012900
		XPLOT(KKK)=DWORKS(LML+1)		00012910
15		YPLOT(KKK)=DWORKS(LML+2)		00012920
		NCFS=NCFS2		00012930
16		CALL SORT(YPLOT,XPLOT,NCFS,BETH,BETL,DWKSP2)		00012940
		IF(NCFS.EQ.0) GO TO 18		00012950
		CALL SORT(XPLOT,YPLOT,NCFS,AHIGH,ALOW,DWKSP2)		00012960
		IF(NCFS.EQ.0) GO TO 18		00012970
		CALL AREA(XPLOT,YPLOT,NCFS,YINTG)		00012980
		SWEET=SWEET+YINTG		00012990
		NSURFI=NSURFI+1		00013000
		NINP(NSURFI)=NCFS		00013010
		N2=2*NCFS		00013020
		DO 17 L=1,NCFS		00013030
		J2=(L-1)*2+1		00013040
		RPSINP(NCUMI+J2)=XPLOT(L)		00013050
17		RPSINP(NCUMI+J2+1)=YPLOT(L)		00013060
		NCUMI=NCUMI+N2		00013070
18		CONTINUE		00013080
		VPROBL=0,		00013090
		IF(NSURFI.EQ.0) RETURN		00013100
		NPTS=NINP(1)		00013110
		N2=2*NPTS		00013120
		DO 30 L=1,N2		00013130
30		RPSOUT(L)=RPSINP(L)		00013140
		NOOUT(1)=NPTS		00013150
		NSURFD=1		00013160
				00013230
				00013240
				00013250
				00013260
				00013270
				00013280

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NCUM0=N2	00013290
IF(NSURFI.EQ.1) GO TO 85	00013300
NCUM1=N2	00013310
DO 60 I=2,NSURFI	00013320
NR=NINP(I)	00013330
N2=2+NR	00013340
DO 40 L=1,N2	00013350
40 RPSWK1(L)=RPSINP(L+NCUM1)	00013360
NWK1(I)=NR	00013370
NSWK1=1	00013380
NCUM1=NCUM1+N2	00013390
NSUM=0	00013400
DO 70 K=1,NSURFO	00013410
NS=NOUT(K)	00013420
42=2+NS	00013430
DO 50 L=1,M2	00013440
50 SPS(L)=RPSOUT(NSUM+L)	00013450
NSUM=NSUM+M2	00013460
NCUM1=0	00013470
NCUM3=0	00013480
NSWK3=0	00013490
DO 62 J=1,NSWK1	00013500
NR=NWK1(J)	00013510
N2=2+NR	00013520
DO 55 L=1,N2	00013530
55 RPS(L)=RPSWK1(L+NCUM1)	00013540
NCUM1=NCUM1+N2	00013570
CALL OVRLAP(RPS, NR, SPS, NS, RPSWK2, NSWK2, NWK2, RPSN)	00013610
IF(NSWK2.EQ.0) GO TO 80	00013620
DO 550 NI=1,NSWK2	00013630
NPTS=NWK2(NI)	00013640
IF(NPTS.GT.50) GO TO 80	00013650
550 CONTINUE	00013660
NCUM2=0	00013670
DO 60 NI=1,NSWK2	00013680
NSWK3=NSWK3+1	00013690
NPTS=NWK2(NI)	00013700
NWK3(NSWK3)=NPTS	00013710
N2=2+NPTS	00013720
DO 56 L=1,N2	00013730
56 RPSWK3(NCUM3+L)=RPSWK2(NCUM2+L)	00013740
NCUM2=NCUM2+N2	00013750
NCUM3=NCUM3+N2	00013760
60 CONTINUE	00013770
62 CONTINUE	00013780
NSWK1=NSWK3	00013790
NCUM1=0	00013800
DO 68 J=1,NSWK1	00013810
NPTS=NWK3(J)	00013820
N2=2+NPTS	00013830
NWK1(J)=NPTS	00013840
DO 65 L=1,N2	00013850

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65	RPSWK1(NCUM1+L)=RPSWK3(NCUM1+L)	00013860
68	NCUM1=NCUM1+N2	00013870
70	CONTINUE	00013880
	NCUM1=0	00013890
	DO 74 J=1,NSWK1	00013900
	NPTS=NWK1(J)	00013910
	NOUT(NSURFO+J)=NPTS	00013920
	N2=2*NPTS	00013930
	DO 72 L=1,N2	00013940
72	RPSOUT(NSUM+L)=RPSWK1(NCUM1+L)	00013950
	NCUM1=NCUM1+N2	00013960
	NSUM=NSUM+N2	00013970
74	CONTINUE	00013980
	NSURFO=NSURFO+NSWK1	00013990
80	CONTINUE	00014000
85	CONTINUE	00014010
	NCUM0=0	00014120
	SWA2=0	00014150
	DO 95 I=1,NSURFO	00014160
	NPTS=NOUT(I)	00014170
	N2=2*NPTS	00014180
	DO 93 L=1,NPTS	00014200
	L2=(L-1)*2+1	00014210
	XPLOT(L)=RPSOUT(NCUM0+L2)	00014220
93	YPLOT(L)=RPSOUT(NCUM0+L2+1)	00014230
	CALL AREA(XPLOT,YPLOT,NPTS,YINTG)	00014250
	SWA2=SWA2+YINTG	00014280
95	NCUM0=NCUM0+N2	00014290
	VPROBL=SWA2/TAREA	00014330
	IF(SWA2.GT.SLEEP) VPROBL=SLEEP/TAREA	00014340
	IF(VPROBL.LE.1.) RETURN	00014341
	WRITE(6,200)	00014342
200	FORMAT(1H0, "WARNING PROGRAM HAD DIFFICULTY ACCOUNTING FOR")	00014343
	1* OVERLAP PROBABILITY GREATER THAN ONE, ONE ASSUMED*	00014344
	VPROBL=1.	00014345
	RETURN	00014350
	END	00014360

6 LEVEL	21	OVERLAP	DATE = 78102	10/47/48
		SUBROUTINE OVRLAP(RPS,NR,SPS,NS,RPSIN,MRPLSN,NRPTSN,RPSN)		00014370
		DIMENSION RPS(1),SPS(1),RPSN(1),RPSIN(1),NRPTSN(1)		00014380
		CALL CHECK4(RPS,NR,SPS,NS)		00014390
		NRPLSN=0		00014400
		NSUM=0		00014410
		IR=1		00014420
		IR2=1		00014430
16		XC=RPS(IR2)		00014440
		YC=RPS(IR2+1)		00014450
		NS2=2*NS		00014460
		CALL CIRCLE(SPS,NS,XC,YC,INCR)		00014470
		IF(INCR,NE,2) GO TO 17		00014480
		IF(IR.EQ.NR) GO TO 120		00014490
		IR=IR+1		00014500
		IR2=2*IR-1		00014510
		GO TO 16		00014520
17		IFLAG=5		00014530
		IF(INCR.EQ.1) IFLAG=6		00014540
		NEXT=0		00014550
		IF(IFLAG.EQ.6) NRN=0		00014560
		IF(IFLAG.EQ.6) GO TO 25		00014570
		NRPLSN=NRPLSN+1		00014580
		NRN=1		00014590
		RPSN(1)=XC		00014600
		RPSN(2)=YC		00014610
		IFLAG=1		00014620
25		IRP1=IR+1		00014630
		IF(IR.EQ.NR) IRP1=1		00014640
		IR2P1=2*IRP1-1		00014650
		IDOUBL=0		00014660
30		X1=RPS(IR2)		00014670
		Y1=RPS(IR2+1)		00014680
35		X2=RPS(IR2P1)		00014690
		Y2=RPS(IR2P1+1)		00014700
		CALL INTERC(SPS,NS,X1,Y1,X2,Y2,X,Y,KL,IAULT,IDOUBL,		00014710
1		XALSO,YALSO)		00014720
		IF(IFLAG.EQ.6.AND.IAULT.EQ.0) GO TO 70		00014730
		IF(IFLAG.EQ.1.AND.IAULT.EQ.0) GO TO 40		00014740
		IF(IFLAG.EQ.4.AND.IAULT.EQ.0) GO TO 40		00014750
		IF(IFLAG.EQ.6) GO TO 38		00014760
		CALL CHECK1(X2,Y2,RPSN,NRN,ICHK)		00014770
		IF(ICHK.EQ.1) GO TO 90		00014780
		IFLAG=1		00014790
		NRN=NRN+1		00014800
		NRN2=2*NRN-1		00014810
		RPSN(NRN2)=X2		00014820
		RPSN(NRN2+1)=Y2		00014830
38		IR=IRP1		00014840
		IR2=2*IR-1		00014850
		IF(IRP1.EQ.1.AND.IFLAG.EQ.6) GO TO 90		00014860
		IRP1=IR+1		00014870
		IF(IR.EQ.NR) IRP1=1		00014880

G LEVEL	21	OVR LAP	DATE = 78102	10/47/48
	IR2P1=2*IRP1-1		00014890	
	IDOUBL=0		00014900	
	GO TO 30		00014910	
40	IFLAG=2		00014920	
	NEXT=IRP1		00014930	
	XSEP=X		00014940	
	YSEP=Y		00014950	
	CALL CHECK1(X,Y,RPSN,NRN,ICHK)		00014960	
	IF(ICHK.EQ.1) GO TO 90		00014970	
	NRN=NRN+1		00014980	
	NRN2=2*NRN-1		00014990	
	RPSN(NRN2)=X		00015000	
	RPSN(NRN2+1)=Y		00015010	
	IS=KL		00015020	
	IS2=2*IS-1		00015030	
	ISP1=IS+1		00015040	
	IF(IS.EQ.NS) ISP1=1		00015050	
	IS2P1=2*ISP1-1		00015060	
	IDOUBL=1		00015070	
	XALSO=X		00015080	
	YALSO=Y		00015090	
	KDIR=0		00015100	
50	X1=SPS(IS2)		00015110	
	Y1=SPS(IS2+1)		00015120	
	X2=SPS(IS2P1)		00015130	
	Y2=SPS(IS2P1+1)		00015140	
	CALL INTER(RPS,NR,X1,Y1,X2,Y2,X,Y,KL,FAULT,IDOUBL,XALSO,YALSO)		00015150	
	IF(IFLAG.EQ.2.AND.IFAULT.EQ.0) GO TO 75		00015160	
	IF(IFLAG.EQ.3.AND.IFAULT.EQ.0) GO TO 75		00015170	
	IF(IFLAG.EQ.3.AND.KDIR.EQ.1) GO TO 60		00015180	
	IF(IFLAG.EQ.3) GO TO 56		00015190	
	IF(IFLAG.EQ.7.AND.IFAULT.EQ.0) GO TO 75		00015200	
	IF(IFLAG.EQ.7.AND.KDIR.EQ.1) GO TO 54		00015210	
	CALL CIRCLE(RPS,NR,X1,Y1,INCR)		00015220	
	IF(INCR.EQ.1) GO TO 50		00015230	
	IF(IFLAG.EQ.1) GO TO 30		00015240	
54	CALL CIRCLE(RPS,NR,X2,Y2,INCR)		00015250	
	IF(INCR.EQ.1) GO TO 60		00015260	
	IF(IFLAG.EQ.7) IFLAG=1		00015270	
	IF(IFLAG.EQ.1) GO TO 30		00015280	
	IF(IFLAG.EQ.7) IFLAG=1		00015290	
	IFLAG=7		00015300	
	IDOUBL=1		00015310	
	XALSO=X		00015320	
	YALSO=Y		00015330	
	IF(INCR.EQ.0) GO TO 58		00015340	
	GO TO 68		00015350	
56	NRN=NRN+1		00015360	
	NRN2=2*NRN-1		00015370	
	RPSN(NRN2)=X1		00015380	
	RPSN(NRN2+1)=Y1		00015390	
	IDOUBL=0		00015400	

G LEVEL	21	OVERLAP	DATE = 78102	10/47/48
	IFLAG=3			00015410
58	ISP1=IS			00015420
	IS2P1=2*ISP1-1			00015430
	IS=IS-1			00015440
	IF (IS.EQ.0) IS=NS			00015450
	IS2=2*IS-1			00015460
	KDIR=-1			00015470
	GO TO 50			00015480
60	NRN=NRN+1			00015490
	NRN2=2*NRN-1			00015500
	RPSN(NRN2)=X2			00015510
	RPSN(NRN2+1)=Y2			00015520
	IDOUBLE=0			00015530
	IFLAG=3			00015540
68	IS=ISP1			00015550
	IS2=2*IS-1			00015560
	ISP1=IS+1			00015570
	IF (IS.EQ.NS) ISP1=1			00015580
	IS2P1=2*ISP1-1			00015590
	KDIR=1			00015600
	GO TO 50			00015610
70	IF (NRPLSN.EQ.0) GO TO 74			00015620
	CALL CHECK2(X,Y,RPSIN,NRPLSN,NRPTSN,ICHK2)			00015630
	IF (ICHK2.EQ.1) GO TO 120			00015640
74	NRPLSN=NRPLSN+1			00015650
	GO TO 76			00015660
75	CALL CHECK1(X,Y,RPSN,NRN,ICHK)			00015670
	IF (ICHK.EQ.1) GO TO 90			00015680
76	NRN=NRN+1			00015690
	NRN2=2*NRN-1			00015700
	RPSN(NRN2)=X			00015710
	RPSN(NRN2+1)=Y			00015720
	X1=X			00015730
	Y1=Y			00015740
	IDOUBL=1			00015750
	XALSO=X			00015760
	YALSO=Y			00015770
	IF (IFLAG.EQ.6) IFLAG=1			00015780
	IF (IFLAG.EQ.1) GO TO 35			00015790
	IRP1=KL+1			00015800
	IF (KL.EQ.NR) IRP1=1			00015810
	IR2P1=2*IRP1-1			00015820
	IFLAG=4			00015830
	GO TO 35			00015840
90	IF (NRN.EQ.0) GO TO 120			00015850
	IF (NRN.LE.2) NRPLSN=NRPLSN-1			00015860
	IF (NRN.LE.2) GO TO 120			00015870
	IF (NEXT.EQ.0) GO TO 100			00015880
	NRPTSN(NRPLSN)=NRN			00015890
	N2=2*NRN			00015900
	DO 95 I=1,N2			00015910
95	RPSIN(I+NSUM)=RPSN(I)			00015920

G LEVEL	21	OVRLAP	DATE = 78102	10/47/48
		NSUM=NSUM+N2		00015930
		NRN=0		00015940
		X1=XSEP		00015950
		Y1=YSEP		00015960
		IFLAG=6		00015970
		IRP1=NEXT		00015980
		IR2P1=2+IRP1-1		00015990
		IDOUBLE=1		00016000
		XALSO=XSEP		00016010
		YALSO=YSEP		00016020
		IF(NRPLSN.EQ.5) GO TO 120		00016030
		GO TO 35		00016040
100		X1=SPS(1)		00016050
		Y1=SPS(2)		00016060
		CALL CIRCLE(RPSN, NRN, X1, Y1, INCR)		00016070
		IF(INCR.EQ.0) GO TO 110		00016080
		X1=RPSN(NRN2)		00016090
		Y1=RPSN(NRN2+1)		00016100
		CALL CLOSE(SPS, NS, X1, Y1, KN)		00016110
		X1=RPSN(1)		00016120
		Y1=RPSN(2)		00016130
		CALL CLOSE(SPS, NS, X1, Y1, K1)		00016140
		DO 103 KDIR=1,2		00016150
		K=KN		00016160
		IFACT=(-1)**(KDIR+1)		00016170
		DO 102 I=1,NS		00016180
		L=K+IFACT*(I-1)		00016190
		IF(L.GT.NS) L=1		00016200
		IF(L.LT.1) L=NS		00016210
		IF(L.EQ.1.AND.KDIR.EQ.1) K=2-I		00016220
		IF(L.EQ.NS.AND.KDIR.EQ.2) K=NS+I-1		00016230
		L2=2+L-1		00016240
		M=NRN+I		00016250
		M2=2+M-1		00016260
		RPSN(M2)=SPS(L2)		00016270
		RPSN(M2+1)=SPS(L2+1)		00016280
		IF(L.EQ.K1) GO TO 1020		00016290
102		CONTINUE		00016300
1020		DO 1025 KK=1,NS		00016310
		KK2=2+KK-1		00016320
		X1=SPS(KK2)		00016330
		Y1=SPS(KK2+1)		00016340
		CALL CIRCLE(RPSN, M, X1, Y1, INCR)		00016350
		IF(INCR.EQ.1) GO TO 103		00016360
1025		CONTINUE		00016370
		GO TO 104		00016380
103		CONTINUE		00016390
		WRITE(6,200)		00016400
200		FORMAT(1H1,*SOMETHING WRONG OVRLAP SUBROUTINE*)		00016410
		STOP		00016420
104		NRPTSN(NRPLSN)=M		00016430
		N2=2+M		00016440

G LEVEL	21	OVERLAP	DATE = 78102	10/47/48
	DO 1040 I=1,N2			00016450
1040	RPSIN(I+NSUM)=RPSN(I)			00016460
	NSUM=NSUM+N2			00016470
	RPSN(1)=RPS(NRN2)			00016480
	RPSN(2)=RPS(NRN2+1)			00016490
	K=KN			00016500
	DO 107 I=1,NS			00016510
	L=K-IFACT*(I-1)			00016520
	IF(L>ST,NS) L=1			00016530
	IF(L<LT,1) L=NS			00016540
	IF(L.EQ.1.AND.IFACT.EQ.-1) K=2-I			00016550
	IF(L.EQ.NS.AND.IFACT.EQ.1) K=NS+I-1			00016560
	L2=2*L-1			00016570
	M=I+1			00016580
	M2=2*M-1			00016590
	RPSN(M2)=SPS(L2)			00016600
	RPSN(M2+1)=SPS(L2+1)			00016610
	IF(L.EQ.K) GO TO 108			00016620
107	CONTINUE			00016630
108	M=M+1			00016640
	M2=2*M-1			00016650
	RPSN(M2)=RPS(1)			00016660
	RPSN(M2+1)=RPS(2)			00016670
	NRPLSN=NRPLSN+1			00016680
	GO TO 115			00016690
110	M=NRN			00016700
115	NRPTSN(NRPLSN)=M			00016710
	N2=2*M			00016720
	DO 118 I=1,N2			00016730
118	RPSIN(I+NSUM)=RPSN(I)			00016740
	NSUM=NSUM+N2			00016750
120	CONTINUE			00016760
	RETURN			00016770
	END			00016780

G LEVEL 21	INTERC	DATE = 78102	10/47/48
SUBROUTINE INTERC(POINTS,NPTS,X1,Y1,X2,Y2,X,Y,KL,IAFAULT,			00016790
1 IDOUBL,XALSO,YALSO)			00016800
DIMENSION POINTS(1)			00016810
DIMENSION IPTS(200),XINT(200),YINT(200)			00016820
COMMON/TOLRS/TOLEQN,TOLSNT,TOLGPT,TOL+TOLCK1,TOLCK2,			00016830
1 TOLCK4,DELT,TOLCIR,ANGST2,TOLPLT			00016840
NINTC=0			00016850
IAFAULT=1			00016860
DO 74 I=1,NPTS			00016870
IP1=I+1			00016880
IF(I.EQ.NPTS) IP1=1			00016890
I2=2*I-1			00016900
I2P1=2*IP1-1			00016910
X3=POINTS(I2)			00016920
Y3=POINTS(I2+1)			00016930
X4=POINTS(I2P1)			00016940
Y4=POINTS(I2P1+1)			00016950
XMIN=X3			00016960
XMAX=X3			00016970
YMIN=Y3			00016980
YMAX=Y3			00016990
IF(X4.LT.X3) XMIN=X4			00017000
IF(X4.GT.X3) XMAX=X4			00017010
IF(Y4.LT.Y3) YMIN=Y4			00017020
IF(Y4.GT.Y3) YMAX=Y4			00017030
XMIN=XMIN-TOL/2.			00017040
YMIN=YMIN-TOL/2.			00017050
XMAX=XMAX+TOL/2.			00017060
YMAX=YMAX+TOL/2.			00017070
DEL1=X2-X1			00017080
DEL2=X4-X3			00017090
IF(ABS(DEL1).LT.TOL) GO TO 10			00017100
SA1=(Y2-Y1)/DEL1			00017110
SB1=Y1-SA1*X1			00017120
10 IF(ABS(DEL2).LT.TOL) GO TO 20			00017130
SA2=(Y4-Y3)/DEL2			00017140
SB2=Y3-SA2*X3			00017150
20 IF(ABS(DEL1).GE.TOL) GO TO 40			00017160
IF(ABS(DEL2).GE.TOL) GO TO 30			00017170
GO TO 74			00017180
30 NINTC=NINTC+1			00017190
XINT(NINTC)=X1			00017200
YINT(NINTC)=SA2*X1+SB2			00017210
IPTS(NINTC)=I			00017220
GO TO 70			00017230
40 IF(ABS(DEL2).GE.TOL) GO TO 50			00017240
NINTC=NINTC+1			00017250
XINT(NINTC)=X3			00017260
YINT(NINTC)=SA1*X3+SB1			00017270
IPTS(NINTC)=I			00017280
GO TO 70			00017290
50 DIFFA=SA1-SA2			00017300

G LEVEL 21	INTERC	DATE = 78102	10/47/48
IF(ABS(DIFFA).LT.TOL) GO TO 74		00017310	
NINTC=NINTC+1		00017320	
XINT(NINTC)=(SB2-SB1)/DIFFA		00017330	
YINT(NINTC)=SA1*XINT(NINTC)+SB1		00017340	
IPTS(NINTC)=I		00017350	
70 CONTINUE		00017360	
CALL CHECK3(XMIN,XMAX,YMIN,YMAX,XINT(NINTC),YINT(NINTC),IBETW)		00017370	
IF(IBETW.EQ.1) IPTS(NINTC)=0		00017380	
74 CONTINUE		00017390	
IF(NINTC.GT.0) IFAULT=0		00017400	
IF(IFault.EQ.1) RETURN		00017410	
XMIN=X1		00017420	
XMAX=X1		00017430	
YMIN=Y1		00017440	
YMAX=Y1		00017450	
IF(X2.LT.X1) XMIN=X2		00017460	
IF(X2.GT.X1) XMAX=X2		00017470	
IF(Y2.LT.Y1) YMIN=Y2		00017480	
IF(Y2.GT.Y1) YMAX=Y2		00017490	
XMIN=XMIN-TOL/2.		00017500	
YMIN=YMIN-TOL/2.		00017510	
XMAX=XMAX+TOL/2.		00017520	
YMAX=YMAX+TOL/2.		00017530	
DO 80 I=1,NINTC		00017540	
IF(IDOUBL.EQ.0) GO TO 75		00017550	
IF(ABS(XINT(I)-XALSO).LT.TOL.AND.ABS(YINT(I)-YALSO).LT.		00017560	
1 TOL) IPTS(I)=0		00017570	
75 IF(XINT(I).LT.XMIN) IPTS(I)=0		00017580	
IF(XINT(I).GT.XMAX) IPTS(I)=0		00017590	
IF(YINT(I).LT.YMIN) IPTS(I)=0		00017600	
IF(YINT(I).GT.YMAX) IPTS(I)=0		00017610	
80 CONTINUE		00017620	
IFFAULT=1		00017630	
DO 90 I=1,NINTC		00017640	
IF(IPTS(I).EQ.0) GO TO 90		00017650	
IF(IFault.EQ.0) GO TO 85		00017660	
IFault=0		00017670	
DIST=(XINT(I)-X1)**2+(YINT(I)-Y1)**2		00017680	
M=I		00017690	
GO TO 90		00017700	
85 DIST2=(XINT(I)-X1)**2+(YINT(I)-Y1)**2		00017710	
IF(DIST2.LT.DIST) M=I		00017720	
IF(M.EQ.I) DIST=DIST2		00017730	
90 CONTINUE		00017740	
IF(IFault.EQ.1) RETURN		00017750	
KL=IPTS(M)		00017760	
X=XINT(M)		00017770	
Y=YINT(M)		00017780	
RETURN		00017790	
END		00017800	

G LEVEL	21	CLOSE	DATE = 78102	10/47/48
SUBROUTINE CLOSE(ARRAY,NPTS,X1,Y1,K)				
DIMENSION ARRAY(1)				
K=1				
DIST=(ARRAY(1)-X1)**2+(ARRAY(2)-Y1)**2				
IF(NPTS.EQ.1) RETURN				
DO 10 I=2,NPTS				
I2=2+I-1				
DIST2=(ARRAY(I2)-X1)**2+(ARRAY(I2+1)-Y1)**2				
IF(DIST2.LT.DIST) K=I				
10 CONTINUE				
RETURN				
END				

G LEVEL 21	CHECK1	DATE = 78102	10/47/48
SUBROUTINE CHECK1(X1,Y1,ARRAY,N,ICOMP)			00017930
DIMENSION ARRAY(1)			00017940
COMMON/TOLRS/TOLFQN,TOLSNT,TOLOPT,TOLINT,TOL,TOLCK2,			00017950
1 TOLCK4,DELT,TOLCIR,ANGST2,TOLPLT			00017960
ICOMP=0			00017970
DO 10 I=1,N			00017980
I2=2*I-1			00017990
X=ARRAY(I2)			00018000
Y=ARRAY(I2+1)			00018010
IF(ABS(X-X1).LT.TOL.AND.ABS(Y-Y1).LT.TOL) GO TO 15			00018020
10 CONTINUE			00018030
RETURN			00018040
15 ICOMP=1			00018050
RETURN			00018060
END			00018070

G LEVEL 21

CHECK2

DATE = 78102

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SUBROUTINE CHECK2(X1,Y1,ARRAY,NPLS,NPTAR,ICOMP)	00018080
DIMENSION ARRAY(1),NPTAR(1)	00018090
COMMON/TOLRS/TOLEQN,TOLSNT,TOLQPT,TOLINT,TOLCK1,TOL,	00018100
1 TOLCK4,DELT,TOLCIR,ANGST2,TOLPLT	00018110
ICOMP=0	00018120
NSUM=0	00018130
DO 20 L=1,NPLS	00018140
NPTS=NPTAR(L)	00018150
DO 10 I=1,NPTS	00018160
K2=NSUM+2*I-1	00018170
X=ARRAY(K2)	00018180
Y=ARRAY(K2+1)	00018190
IF(ABS(X1-X).GE.TOL) GO TO 10	00018200
IF(ABS(Y1-Y).GE.TOL) GO TO 10	00018210
ICOMP=1	00018220
RETURN	00018230
10 CONTINUE	00018240
NSUM=NSUM+2*NPTS	00018250
20 CONTINUE	00018260
RETURN	00018270
END	00018280

G LEVEL 21	CHECK3	DATE = 78102	10/47/48
SUBROUTINE CHECK3(XMIN,XMAX,YMIN,YMAX,X,Y,IBETW)			00018290
IBETW=0			00018300
IF(X.LT.XMIN) IBETW=1			00018310
IF(X.GT.XMAX) IBETW=1			00018320
IF(Y.LT.YMIN) IBETW=1			00018330
IF(Y.GT.YMAX) IBETW=1			00018340
RETURN			00018350
END			00018360

G LEVEL	21	CHECK#	DATE = 78102	10/47/48
		SUBROUTINE CHECK4(RPS,NR,SPS,NS)		00018370
		DIMENSION RPS(1),SPS(1)		00018380
		COMMON/TOLRS/TOLEQM,TOLSNT,TOLOPT,TOLINT,TOLCK1,TOLCK2,		00018390
1		TOL,DELT,TOLCIR,ANGST2,TOLPLT		00018400
		DO 30 I=1,MR		00018410
		I2=(I-1)*2+1		00018420
		I2P2=I2+2		00018430
		IF(I.EQ.NR) I2P2=1		00018440
		XR1=RPS(I2)		00018450
		YR1=RPS(I2+1)		00018460
		XR2=RPS(I2P2)		00018470
		YR2=RPS(I2P2+1)		00018480
		ISWTCH=1		00018490
		IF(ABS(XR2-XR1).LE.TOL) ISWTCH=2		00018500
		IF(ABS(YR2-YR1).LE.TOL) ISWTCH=3		00018510
		IF(ISWTCH.NE.1) GO TO 15		00018520
		ASLOPE=(YR2-YR1)/(XR2-XR1)		00018530
		BINT=YR1-ASLOPE*XR1		00018540
		AINSL=-1./ASLOPE		00018550
15		DO 20 J=1,NS		00018560
		J2=(J-1)*2+1		00018570
		XS1=SPS(J2)		00018580
		YS1=SPS(J2+1)		00018590
		GO TO (16,17,18), ISWTCH		00018600
16		XS1L=(YR1-YS1*AINSL*XS1-ASLOPE*XR1)/(AINSL-ASLOPE)		00018610
		YS1L=YR1-ASLOPE*(XS1L-XR1)		00018620
		DIST=SQRT((XS1L-XS1)**2+(YS1L-YS1)**2)		00018630
		IF(DIST.GT.TOL) GO TO 20		00018640
		ALPHA=ATAN(AINSL)		00018650
		XS1P=XS1L*DELT*COS(ALPHA)		00018660
		YS1P=YS1L*DELT*SIN(ALPHA)		00018670
		CALL CIRCLE(RPS,NR,XS1P,YS1P,INCR)		00018680
		IF(INCR.EQ.0) GO TO 160		00018690
		XS1P=XS1L-DELT*COS(ALPHA)		00018700
		YS1P=YS1L-DELT*SIN(ALPHA)		00018710
160		SPS(J2)=XS1P		00018720
		SPS(J2+1)=YS1P		00018730
		GO TO 20		00018740
17		IF(ABS(XS1-XR2).GT.TOL) GO TO 20		00018750
		XS1P=XS1*DELT		00018760
		YS1P=YS1		00018770
		CALL CIRCLE(RPS,NR,XS1P,YS1P,INCR)		00018780
		IF(INCR.EQ.0) GO TO 160		00018790
		XS1P=XS1-DELT		00018800
		GO TO 160		00018810
18		IF(ABS(YS1-YR2).GT.TOL) GO TO 20		00018820
		YS1P=YS1*DELT		00018830
		XS1P=XS1		00018840
		CALL CIRCLE(RPS,NR,XS1P,YS1P,INCR)		00018850
		IF(INCR.EQ.0) GO TO 160		00018860
		YS1P=YS1-DELT		00018870
		GO TO 160		00018880

G LEVEL 21

CHECK4

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20 CONTINUE
30 CONTINUE
RETURN
END

00018890
00018900
00018910
00018920

G LEVEL 21	CIRCLE	DATE = 78102	10/47/48
SUBROUTINE CIRCLE(POINTS,NPTS,XC,YC,INCR)		00018930	
DIMENSION POINTS(1)		00018940	
COMMON/TOLRS/TOLEQN,TOLSNT,TOLQPT,TOLINT,TOLCK1,TOLCK2,		00018950	
1 TOLCK4,DELT,TOL1,ANGST2,TOLPLT		00018960	
NPC=0		00018970	
NNC=0		00018980	
DO 10 I=1,NPTS		00018990	
IP1=I+1		00019000	
IF(I.EQ.NPTS) IP1=1		00019010	
L=(I-1)*2		00019020	
LP1=(IP1-1)*2		00019030	
DISX1=POINTS(L+1)-XC		00019040	
DISY1=POINTS(L+2)-YC		00019050	
DISX2=POINTS(LP1+1)-XC		00019060	
DISY2=POINTS(LP1+2)-YC		00019070	
IF(ABS(DISX1).LT.TOL1.AND.ABS(DISY1).LT.TOL1) GO TO 20		00019080	
IF(ABS(DISX2).LT.TOL1.AND.ABS(DISY2).LT.TOL1) GO TO 20		00019090	
IF(ABS(DISX1).GE.TOL1) GO TO 3		00019100	
IF(ABS(DISX2).GE.TOL1) GO TO 1		00019110	
IF((DISY1/DISY2).GT.0.0) GO TO 10		00019120	
GO TO 20		00019130	
1 IF(ABS(DISY2).GE.TOL1) GO TO 2		00019140	
X=DISX2		00019150	
GO TO 8		00019160	
2 IF(ABS(DISY1-DISY2).LT.TOL1) GO TO 10		00019170	
GO TO 5		00019180	
3 IF(ABS(DISY1).GE.TOL1) GO TO 32		00019190	
IF(ABS(DISX2).LT.TOL1) GO TO 10		00019200	
IF(ABS(DISY2).GE.TOL1) GO TO 10		00019210	
IF((DISX1/DISX2).GT.0.0) GO TO 10		00019220	
GO TO 20		00019230	
32 IF(ABS(DISX2).GE.TOL1) GO TO 34		00019240	
GO TO 36		00019250	
34 IF(ABS(DISY2).GE.TOL1) GO TO 36		00019260	
X=DISX2		00019270	
GO TO 8		00019280	
36 IF(ABS(DISY2-DISY1).LT.TOL1) GO TO 10		00019290	
IF(ABS(DISX2-DISX1).GE.TOL1) GO TO 5		00019300	
IF((DISY1/DISY2).GT.0.0) GO TO 10		00019310	
X=DISX2		00019320	
GO TO 8		00019330	
5 SA=(DISY2-DISY1)/(DISX2-DISX1)		00019340	
SB=DISY1-SA*DISX1		00019350	
X=-SB/SA		00019360	
IF(X.LT.DISX1.AND.X.LT.DISX2) GO TO 10		00019370	
IF(X.GT.DISX1.AND.X.GT.DISX2) GO TO 10		00019380	
8 IF(ABS(X).LT.TOL1) GO TO 20		00019390	
IF(X.LT.0.0) NNC=NNC+1		00019400	
IF(X.GT.0.0) NPC=NPC+1		00019410	
10 CONTINUE		00019420	
INCR=0		00019430	
IF(2*(NPC/2).EQ.NPC) RETURN		00019440	

G LEVEL	21	CIRCLE	DATE = 78102	10/47/48
		IF (2*(NNC/2).EQ.NNC) RETURN		00019450
		INCR=1		00019460
		RETURN		00019470
20		INCR=2		00019480
		RETURN		00019490
		END		00019500

G LEVEL 21	AREA	DATE = 78102	10/47/48
SUBROUTINE AREA(X,Y,NPTS,YINTG)			00019510
DIMENSION X(1),Y(1)			00019520
XLOW=X(1)			00019530
XHIGH=XLOW			00019540
KMIN=1			00019550
KMAX=1			00019560
DO 10 I=2,NPTS			00019570
IF(X(I).GE.XLOW) GO TO 5			00019580
KMIN=I			00019590
XLOW=X(I)			00019600
5 IF(X(I).LE.XHIGH) GO TO 10			00019610
KMAX=I			00019620
XHIGH=X(I)			00019630
10 CONTINUE			00019640
IF(KMIN.EQ.KMAX) GO TO 70			00019650
L1=KMIN			00019660
L2=KMAX			00019670
IF(KMAX.LT.KMIN) L2=NPTS+KMAX			00019680
NTABS=2-L1			00019690
YINTG=0.			00019700
DO 20 L=1,NTABS			00019710
K1=L1+L-1			00019720
IF(K1.GT.NPTS) K1=K1-NPTS			00019730
K2=L1+L			00019740
IF(K2.GT.NPTS) K2=K2-NPTS			00019750
DX=X(K2)-X(K1)			00019760
20 YINTG=YINTG+(DX/2.)*(Y(K1)+Y(K2))			00019770
L2=KMAX			00019780
IF(KMAX.GT.KMIN) L2=KMAX-NPTS			00019790
NTABS=L1-L2			00019800
DO 30 L=1,NTABS			00019810
K1=L1-L+1			00019820
IF(K1.LT.1) K1=K1+NPTS			00019830
K2=L1-L			00019840
IF(K2.LT.1) K2=K2+NPTS			00019850
DX=X(K2)-X(K1)			00019860
30 YINTG=YINTG-(DX/2.)*(Y(K1)+Y(K2))			00019870
YINTG=ABS(YINTG)			00019880
RETURN			00019890
70 YLOW=Y(1)			00019900
YHIGH=YLOW			00019910
DO 80 I=2,NPTS			00019920
IF(Y(I).LT.YLOW) YLOW=Y(I)			00019930
IF(Y(I).GT.YHIGH) YHIGH=Y(I)			00019940
80 CONTINUE			00019950
YINTG=YHIGH-YLOW			00019960
RETURN			00019970
END			00019980

G LEVEL 21	PLOTOL	DATE = 78102	10/47/48	
	SUBROUTINE PLOTOL(DATAC,DATAS,MS,TITLES,GAMMA,A PLOT,BPLOT,	00019990		
1	VPROBL,IPLTOL)	00020000		
	DIMENSION IPLTOL(1)	00020010		
	DIMENSION DATAC(1),DATAS(1),MS(1),TITLES(1),APLOT(1),BPLOT(1)	00020020		
	DIMENSION XPR(10),OUT(91),SYMBOL(9)	00020030		
	DATA BLANK,PERIOD,FI/1H,1H,1HI/	00020040		
	DATA SYMBOL/1HA,1HB,1HC,1HD,1HE,1HF,1HG,1HK,1HM/	00020050		
	DIMENSION HEAD(9)	00020060		
	WRITE(6,101) TITLES(I),I=1,80)	00020070		
101	FORMAT(1H1,80A1)	00020080		
	WRITE(6,209)	00020090		
	WRITE(6,102) GAMMA,DATAC(4),DATAC(5)	00020100		
	ALOW=DATAC(16)	00020110		
	AHIGH=DATAC(17)	00020120		
	BHIGH=DATAC(19)	00020130		
	BLOW=DATAC(18)	00020140		
	DELX=(BHIGH-BLOW)/90.	00020150		
	DELY=(AHIGH-ALOW)/20.	00020160		
	KPRINT=5	00020170		
	NPANEL=DATAC(23)+0.5	00020180		
	NPSI=DATAC(9)+0.5	00020190		
	DPSI=DATAC(11)	00020200		
	PSI=DATAC(10)-DPSI	00020210		
DO 20	I=1,NPSI	00020220		
	PSI=PSI+DPSI	00020230		
20	HEAD(I)=PSI	00020240		
	TY2=AHIGH+DELY/2.	00020250		
DO 140	I=1,21	00020260		
	TY1=TY2-DELY	00020270		
DO 30	J=1,91	00020280		
30	OUT(J)=BLANK	00020290		
	IF(KPRINT.EQ.5) OUT(1)=FI	00020300		
	IF(KPRINT.NE.5) OUT(1)=PERIOD	00020310		
	IF(KPRINT.EQ.5) OUT(91)=FI	00020320		
	IF(KPRINT.NE.5) OUT(91)=PERIOD	00020330		
	IF(I,NE.1,AND.I,NE.21) GO TO 50	00020340		
	L=9	00020350		
DO 40	J=1,91	00020360		
	L=L+1	00020370		
	OUT(J)=PERIOD	00020380		
40	IF(L.EQ.10) OUT(J)=FI	00020390		
	IF(L.EQ.10) L=0	00020400		
50	CONTINUE	00020410		
	MAS=0	00020420		
DO 90	K=1,NPANEL	00020430		
	DO 60	J=1,NPSI	00020440	
	DO 75	L=1,4	00020450	
	MAS=MAS+1	00020460		
	IF(APLOT(MAS).GT.TY2) GO TO 75	00020470		
	IF(APLOT(MAS).LE.TY1) GO TO 75	00020480		
	IF(IPLTOL(MAS).EQ.1) GO TO 75	00020490		
	TX2=BLOW+DELX/2.	00020500		

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DO 55 M=1,91		00020510	
TX1=TX2+DELX		00020520	
IF(BPLOT(MAS).LT.TX2.AND.BPLOT(MAS).GE.TX1) GO TO 58		00020530	
55 TX2=TX2+DELX		00020540	
58 OUT(M)=SYMBOL(J)		00020550	
75 CONTINUE		00020560	
60 CONTINUE		00020570	
90 CONTINUE		00020580	
YPLOT=TY2-DELY/2.		00020590	
IF(KPRINT.EQ.5) WRITE(6,201) YPLOT,OUT,YPLOT		00020600	
IF(KPRINT.NE.5) WRITE(6,205) OUT		00020610	
IF(KPRINT.EQ.5) KPRINT=0		00020620	
KPRINT=KPRINT+1		00020630	
140 TY2=TY2-DELY		00020640	
XPR(I)=BLW		00020650	
DO 150 I=2,10		00020660	
150 XPR(I)=XPR(I-1)+10.+DELX		00020670	
WRITE(6,204)		00020680	
WRITE(6,202) XPR		00020690	
WRITE(6,207)		00020700	
WRITE(6,217) VPROBL		00020710	
217 FORMAT(1H0,105X,*PROBABILITY:*,F6.3)		00020720	
WRITE(6,210)		00020730	
DO 160 L=1,4		00020740	
K=NPSI+L		00020750	
IF(K.LE.4) WRITE(6,211) SYMBOL(L),HEAD(L),SYMBOL(L+5),HEAD(L+5)		00020760	
IF(K.GT.4.AND.K.LE.9) WRITE(6,212) SYMBOL(L),HEAD(L),SYMBOL(L+5)		00020770	
IF(K.GT.9) WRITE(6,213) SYMBOL(L),SYMBOL(L+5)		00020780	
160 CONTINUE		00020790	
IF(NPSI.LT.5) WRITE(6,215) SYMBOL(5)		00020800	
IF(NPSI.GT.5) WRITE(6,216) SYMBOL(5),HEAD(5)		00020810	
WRITE(6,214)		00020820	
210 FORMAT(//1H .22X,76(**)/1H .22X,* * * ,74X,* * * /1H .22X,* * *,2X,		00020830	
1 *KEY TO PLOT SYMBOLS*,53X,* * * /1H .22X,* * * ,74X,* * * /1H .22X,		00020840	
2 * * *.4X,*SYMBOL*,5X,*A/C HEADING (DEG)*,10X,*SYMBOL*,5X,		00020850	
3 *A/C HEADING (DEG)*,4X,* * *)		00020860	
211 FORMAT(1H .22X,* * *,7X,A1+12X,F7.2,18X,A1+12X,F7.2+9X,* * *)		00020870	
212 FORMAT(1H .22X,* * *,7X,A1+12X,F7.2+18X,A1+28X,* * *)		00020880	
213 FORMAT(1H .22X,* * *,7X,A1+37X,A1+28X,* * *)		00020890	
214 FORMAT(1H .22X,* * *,74X,* * */1H .22X,* * *,74X,* * */1H .22X,76(**))		00020900	
215 FORMAT(1H .22X,* * *,7X,A1,66X,* * *)		00020910	
216 FORMAT(1H .22X,* * *,7X,A1+12X,F7.2+47X,* * *)		00020920	
102 FORMAT(1H0,*SUN ELEVATION*,F12.2/1H ,*A/C PITCH ATTITUDE*, 1 F7.2/1H ,*A/C ROLL ATTITUDE*,F8.2/1H ,*OBSERVER ANGLE*,93X, 2 *OBSERVER ANGLE*/1H .3X,*DEGREES*,100X,*DEGREES*/)		00020930 00020940 00020950	
204 FORMAT()		00020960	
202 FORMAT(1H .9X,10F10.3)		00020970	
207 FORMAT(/,38X,*GLINT AZIMUTH WITH RESPECT TO SUN - DEGREES*)		00020980	
201 FORMAT(1H .F10.3,5X,91A1+1X,F10.3)		00020990	
205 FORMAT(1H .15X,91A1)		00021000	
209 FORMAT(1H0,5DX,*SUN GLINT SIGNATURE*)		00021010	
RETURN		00021020	

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END

00021030

G LEVEL 21	DRPLOT	DATE = 78102	10/47/48
SUBROUTINE DRPLOT(DATAC,DATAS,MS,	GAMMA,ALPHA2,BETA2,	00021040	
1 INTRFL,ISHADW,INTERF,XPLOT,YPLOT,IBUF,JCONT,DWORKS,DWKSP2,VPROBL)	00021050		
DIMENSION DATAC(1),DATAS(1),MS(1),TITLES(1),ALPHA2(1),BETA2(1),	00021060		
1 XPLOT(1),YPLOT(1),IBUF(1),INTRFL(1),ISHADW(1),INTERF(1)	00021070		
DIMENSION DWORKS(1),DWKSP2(1)	00021080		
IZPTS=0	00021090		
ICONT=1	00021100		
JCONT=JCONT+1	00021110		
IRFLTN=DATAC(3)+0.5	00021120		
THETA=DATAC(4)	00021130		
PHI=DATAC(5)	00021140		
NPANEL=DATAC(23)+0.5	00021150		
NPSI=DATAC(9)+0.5	00021160		
IALPL=DATAC(16)+0.5	00021170		
IF(DATAC(16).LT.0.) IALPL=DATAC(16)-0.5	00021180		
IDAL=(DATAC(17)-DATAC(16))/2.+0.5	00021190		
IBET=DATAC(18)+0.5	00021200		
IF(DATAC(18).LT.0.) IBET=DATAC(18)-0.5	00021210		
IDBET=(DATAC(19)-DATAC(18))/6.+0.5	00021220		
BETL=IBET	00021230		
DBT=IDBET	00021240		
ALPL=IALPL	00021250		
DAL=IDAL	00021260		
MAS=0	00021270		
DPSI=DATAC(11)	00021280		
DO 90 I=1,NPANEL	00021290		
K=MS(I)	00021300		
NPTS=DATAS(K)+0.5	00021310		
PSI=DATAC(10)-DPSI	00021320		
DO 60 J=1,NPSI	00021330		
PSI=PSI+DPSI	00021340		
MPTS=0	00021350		
DO 75 L=1,NPTS	00021360		
MAS=MAS+1	00021370		
IF(IRFLTN.NE.INTRFL(MAS)) GO TO 75	00021380		
MPTS=MPTS+1	00021390		
XPLOT(MPTS)=BETA2(MAS)	00021400		
YPLOT(MPTS)=ALPHA2(MAS)	00021410		
75 CONTINUE	00021420		
IF(MPTS.EQ.0) GO TO 60	00021430		
CALL SORT(XPLOT,YPLOT,MPTS,DATAC(17),DATAC(16),DWKSP2)	00021440		
IF(MPTS.EQ.0) GO TO 60	00021450		
CALL SORT2(XPLOT,YPLOT,MPTS,MPTS1,MPTS2,DWORKS,DWKSP2)	00021460		
IF(MPTS2.EQ.0) GO TO 77	00021470		
CALL SORT(YPLOT,XPLOT,MPTS1,DATAC(19),DATAC(18),DWKSP2)	00021480		
IF(MPTS1.EQ.0) GO TO 20	00021490		
CALL GPLOT(ALPL,DAL,GAMMA,THETA,PHI,PSI,MPTS1,ICONT,XPLOT,YPLOT,	00021500		
1 IBUF,JCONT,BETL,DBT,VPROBL,IZPTS)	00021510		
ICONT=0	00021520		
20 CONTINUE	00021530		
DO 76 KKK=1,MPTS2	00021540		
LML=(KKK-1)*2	00021550		

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XPLOT(KKK)=DWORKS(LML+1)	00021560
76 YPLOT(KKK)=DWORKS(LML+2)	00021570
MPTS=MPTS2	00021580
77 CONTINUE	00021590
CALL SORT(YPLOT,XPLOT,MPTS,DATA(19),DATA(18),DWKSP2)	00021600
IF(MPTS.EQ.0) GO TO 60	00021610
CALL GPLOT(ALPL,DAL,GAMMA,THETA,PHI,PSI,MPTS,ICONT,XPLOT,YPLOT,	00021620
1 IBUF,JCONT,BETL,DBT,VPROBL,IZPTS)	00021630
ICONT=0	00021640
60 CONTINUE	00021650
90 CONTINUE	00021660
IF(ICONT.EQ.0) RETURN	00021670
IZPTS=1	00021680
CALL GPLOT(ALPL,DAL,GAMMA,THETA,PHI,PSI,MPTS,ICONT,XPLOT,YPLOT,	00021690
1 IBUF,JCONT,BETL,DBT,VPROBL,IZPTS)	00021700
RETURN	00021710
END	00021720

6 LEVEL	21	SORT	DATE = 78102	10/47/48
		SUBROUTINE SORT(XPLOT,YPLOT,MPTS,AHIGH,ALOW,DWORKS)	00021730	
		DIMENSION XPLOT(1),YPLOT(1),DWORKS(1)	00021740	
		INTC=0	00021750	
		LPTS=0	00021760	
		DO 5 I=1,MPTS	00021770	
		KPT=(I-1)*2	00021780	
		DWORKS(KPT+1)=XPLOT(I)	00021790	
5		DWORKS(KPT+2)=YPLOT(I)	00021800	
		DO 70 I=1,MPTS	00021810	
		IP1=I+1	00021820	
		IF(I.EQ.MPTS) IP1=1	00021830	
		IFL1=2	00021840	
		IFL2=2	00021850	
		KPT=(I-1)*2	00021860	
		KPT1=(IP1-1)*2	00021870	
		X1=DWORKS(KPT+1)	00021880	
		Y1=DWORKS(KPT+2)	00021890	
		X2=DWORKS(KPT1+1)	00021900	
		Y2=DWORKS(KPT1+2)	00021910	
		IF(Y1.GT.AHIGH) IFL1=1	00021920	
		IF(Y2.GT.AHIGH) IFL2=1	00021930	
		IF(Y1.LT.ALLOW) IFL1=3	00021940	
		IF(Y2.LT.ALLOW) IFL2=3	00021950	
		IF(Y1.NE.Y2) RATIO=(X2-X1)/(Y2-Y1)	00021960	
		GO TO (1,2,3), IFL1	00021970	
1		GO TO (11,12,13), IFL2	00021980	
2		GO TO (21,22,23), IFL2	00021990	
3		GO TO (31,32,33), IFL2	00022000	
11		IF(INTC.EQ.0) GO TO 60	00022010	
		LPTS=LPTS+1	00022020	
		XPLOT(LPTS)=XSAVE	00022030	
		YPLOT(LPTS)=YSAVE	00022040	
		INTC=0	00022050	
		GO TO 60	00022060	
12		IF(INTC.EQ.0) GO TO 120	00022070	
		LPTS=LPTS+1	00022080	
		XPLOT(LPTS)=XSAVE	00022090	
		YPLOT(LPTS)=YSAVE	00022100	
		INTC=0	00022110	
120		LPTS=LPTS+1	00022120	
		XPLOT(LPTS)=X1+RATIO*(AHIGH-Y1)	00022130	
		YPLOT(LPTS)=AHIGH	00022140	
		GO TO 60	00022150	
13		IF(INTC.EQ.0) GO TO 130	00022160	
		LPTS=LPTS+1	00022170	
		XPLOT(LPTS)=XSAVE	00022180	
		YPLOT(LPTS)=YSAVE	00022190	
130		LPTS=LPTS+1	00022200	
		XPLOT(LPTS)=X1+RATIO*(AHIGH-Y1)	00022210	
		YPLOT(LPTS)=AHIGH	00022220	
		XSAVE=X1+RATIO*(ALOW-Y1)	00022230	
		YSAVE=ALOW	00022240	

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	INTC=1	00022250
	GO TO 60	00022260
21	IF(INTC.EQ.0) GO TO 210	00022270
	LPTS=LPTS+1	00022280
	XPLOT(LPTS)=XSAVE	00022290
	YPLT(LPTS)=YSAVE	00022300
210	LPTS=LPTS+1	00022310
	XPLT(LPTS)=X1	00022320
	YPLT(LPTS)=Y1	00022330
	XSAVE=X1+RATIO*(AHIGH-Y1)	00022340
	YSAVE=AHIGH	00022350
	INTC=1	00022360
	GO TO 60	00022370
22	IF(INTC.EQ.0) GO TO 220	00022380
	LPTS=LPTS+1	00022390
	XPLT(LPTS)=XSAVE	00022400
	YPLT(LPTS)=YSAVE	00022410
	INTC=0	00022420
220	LPTS=LPTS+1	00022430
	XPLT(LPTS)=X1	00022440
	YPLT(LPTS)=Y1	00022450
	GO TO 60	00022460
23	IF(INTC.EQ.0) GO TO 230	00022470
	LPTS=LPTS+1	00022480
	XPLT(LPTS)=XSAVE	00022490
	YPLT(LPTS)=YSAVE	00022500
230	LPTS=LPTS+1	00022510
	XPLT(LPTS)=X1	00022520
	YPLT(LPTS)=Y1	00022530
	XSAVE=X1+RATIO*(ALOW-Y1)	00022540
	YSAVE=ALOW	00022550
	INTC=1	00022560
	GO TO 60	00022570
31	IF(INTC.EQ.0) GO TO 310	00022580
	LPTS=LPTS+1	00022590
	XPLT(LPTS)=XSAVE	00022600
	YPLT(LPTS)=YSAVE	00022610
310	LPTS=LPTS+1	00022620
	XPLT(LPTS)=X1+RATIO*(ALOW-Y1)	00022630
	YPLT(LPTS)=ALOW	00022640
	XSAVE=X1+RATIO*(AHIGH-Y1)	00022650
	YSAVE=AHIGH	00022660
	INTC=1	00022670
	GO TO 60	00022680
32	IF(INTC.EQ.0) GO TO 320	00022690
	LPTS=LPTS+1	00022700
	XPLT(LPTS)=XSAVE	00022710
	YPLT(LPTS)=YSAVE	00022720
	INTC=0	00022730
320	LPTS=LPTS+1	00022740
	XPLT(LPTS)=X1+RATIO*(ALOW-Y1)	00022750
	YPLT(LPTS)=ALOW	00022760

G LEVEL	21	SORT	DATE = 78102	10/47/48
	GO TO 60			00022770
33	IF(INTC.EQ.0) GO TO 60			00022780
	LPTS=LPTS+1			00022790
	XPLOT(LPTS)=XSAVE			00022800
	YPLOT(LPTS)=YSAVE			00022810
	INTC=0			00022820
60	IF(INTC.EQ.0) GO TO 70			00022830
	IF(I.LT.MPTS) GO TO 70			00022840
	LPTS=LPTS+1			00022850
	XPLOT(LPTS)=XSAVE			00022860
	YPLOT(LPTS)=YSAVE			00022870
70	CONTINUE			00022880
	MPTS=LPTS			00022890
	RETURN			00022900
	END			00022910

G LEVEL	21	SORT2	DATE = 78102	10/47/48
		SUBROUTINE SORT2(XPLOT,YPLOT,MPTS,MPTS1,MPTS2,DWORKS,DWKSP2)	00022920	
		DIMENSION XPLOT(1),YPLOT(1),DWORKS(1),DWKSP2(1)	00022930	
		COMMON/TDLRS/TOLEQN,TOLSNT,TOLAPT,TOLINT,TOLCK1,TOLCK2,	00022940	
		1 TOLCK4,DELT,TOLCIR,ANGST2,TOLPLT	00022950	
		MPTS2=0	00022960	
		IF(MPTS.EQ.1) RETURN	00022970	
		BMIN=XPLOT(1)	00022980	
		BMAX=BMIN	00022990	
		DO 62 I=2,MPTS	00023000	
		IF(XPLOT(I).LT.BMIN) BMIN=XPLOT(I)	00023010	
		IF(XPLOT(I).GT.BMAX) BMAX=XPLOT(I)	00023020	
	62	CONTINUE	00023030	
		IF(BMIN.GE.0.0) RETURN	00023040	
		IF(BMAX.LE.0.0) RETURN	00023050	
		DO 64 I=1,MPTS	00023060	
		IF(ABS(XPLOT(I)).LE.45.) RETURN	00023070	
	64	CONTINUE	00023080	
		BAVG=(BMAX-BMIN)/2.	00023090	
		IF(BAVG.LT.ANGST2) RETURN	00023100	
		DO 30 I=1,MPTS	00023110	
		LPS=(I-1)*2	00023120	
		DWORKS(LPS+1)=XPLOT(I)	00023130	
		DWORKS(LPS+2)=YPLOT(I)	00023140	
		IF(XPLOT(I).LT.0.0) XPLOT(I)=XPLOT(I)+360.	00023150	
	30	CONTINUE	00023160	
		MPTS1=MPTS	00023170	
		CALL SORT(YPLOT,XPLOT,MPTS1,180.,0.,DWKSP2)	00023180	
		DO 35 I=1,MPTS1	00023190	
		LPS=(I-1)*2	00023200	
		DWKSP2(LPS+1)=XPLOT(I)	00023210	
	35	DWKSP2(LPS+2)=YPLOT(I)	00023220	
		DO 40 I=1,MPTS	00023230	
		LPS=(I-1)*2	00023240	
		XPLOT(I)=DWORKS(LPS+1)	00023250	
		YPLOT(I)=DWORKS(LPS+2)	00023260	
		IF(XPLOT(I).GT.0.) XPLOT(I)=XPLOT(I)-360.	00023270	
	40	CONTINUE	00023280	
		DO 45 I=1,MPTS1	00023290	
		LPS=(I-1)*2	00023300	
		DWORKS(LPS+1)=DWKSP2(LPS+1)	00023310	
	45	DWORKS(LPS+2)=DWKSP2(LPS+2)	00023320	
		MPTS2=MPTS1	00023330	
		MPTS1=MPTS	00023340	
		CALL SORT(YPLOT,XPLOT,MPTS1,0.,-180.,DWKSP2)	00023350	
		RETURN	00023360	
		END	00023370	

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SUBROUTINE GPLOT(ALPL,DAL,SUNELV,THAC,PHAC,HEAD,NPTS,ICONT, 00023380
 1 XPLOT,YPLOT, IBUF,JCONT,BETL,DBT,VPROBL,IZPTS) 00023390
DIMENSION XPLOT(1),YPLOT(1),IBUF(1) 00023400
CALL FACTOR(.7857) 00023410
IF(ICONT.EQ.0)GO TO 10 00023420
GO TO(50,20,20,60,20,20,60,20,20,60,20,20) ,JCONT 00023430
50 CALL PLOT(2,+1,+3) 00023440
CALL PLOT(-2,+1,+3) 00023450
CALL PLOT(0,0,-1,+2) 00023460
CALL PLOT(8,8,13,+2) 00023470
CALL PLOT(-2,+13,+2) 00023480
CALL PLOT(-2,-1,+2) 00023490
GO TO 30 00023500
20 CALL PLOT(0,+4,+3) 00023510
GO TO 30 00023520
60 CALL PLOT(12,+8,+3) 00023530
CALL PLOT(-2,-1,+3) 00023540
CALL PLOT(8,8,-1,+2) 00023550
CALL PLOT(8,8,13,+2) 00023560
CALL PLOT(-2,+13,+2) 00023570
CALL PLOT(-2,-1,+2) 00023580
30 CONTINUE 00023590
C***** DRAW AXIS *****
CALL AXIS(0.,0.,16)HOP SERVER AZIMUTH,-16,6,0.,BETL,DBT) 00023600
CALL AXIS(0.,0.,21)ELEVATION OF OBSERVER,21+2.,90.,ALPL,DAL) 00023620
CALL PLOT(0,+1,+3) 00023630
CALL PLOT(6,+1,+2) 00023640
CALL PLOT(6,+0,+3) 00023650
CALL PLOT(6,+2,+2) 00023660
CALL PLOT(0,+2,+2) 00023670
CALL GRID(.01,.01,1.,1.,6,2) 00023680
CALL SYMBOL(5.,+2.3,G+10.+12)PROBABILITY .0..11) 00023690
CALL NUMBER(999.,999.,+0.1D+0)VPROBL,0.,3) 00023700
IF(IZPTS.EQ.1) GO TO 19 00023710
10 CONTINUE 00023720
XPLOT(NPTS+1)=XPLOT(1) 00023730
YPLOT(NPTS+1)=YPLOT(1) 00023740
XPLOT(NPTS+2)=BETL 00023750
XPLOT(NPTS+3)=DBT 00023760
YPLOT(NPTS+2)=ALPL 00023770
YPLOT(NPTS+3)=DAL 00023780
ND=NPTS+1 00023790
CALL LINE(XPLOT,YPLOT,ND,+0,4) 00023800
IF(HEAD.LT.0.) GO TO 15 00023810
AMAX=YPLOT(1) 00023820
LMAX=1 00023830
DO 12 I=2,NPTS 00023840
IF(YPLOT(I).GT.AMAX) LMAX=I 00023850
IF(LMAX.EQ.I) AMAX=YPLOT(LMAX) 00023860
12 CONTINUE 00023870
AMAX=(YPLOT(LMAX)-ALPL)/DAL+0.05 00023880
BMAX=(XPLOT(LMAX)-BETL)/DBT 00023890

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CALL NUMBER(BMAX,AMAX,0.07,HEAD,0.,-1)	00023900
GO TO 19	00023910
15 AMIN=YPLOT(1)	00023920
LMIN=1	00023930
DO 17 I=2,NPTS	00023940
IF(YPLOT(I).LT.AMIN) LMIN=I	00023950
IF(LMIN.EQ.I) AMIN=YPLOT(LMIN)	00023960
17 CONTINUE	00023970
AMIN=(YPLOT(LMIN)-ALPL)/DAL-0.05	00023980
BMIN=(XPLOT(LMIN)-BETL)/DBT	00023990
CALL NUMBER(BMIN,AMIN,0.07,HEAD,0.,-1)	00024000
19 CONTINUE	00024010
IF(ICONT.EQ.0) GO TO 40	00024020
CALL SYMBOL(1.8+2.1,0.10,25HHEADING OF AIRCRAFT (DEG),0.,25)	00024030
CALL SYMBOL(0.,3.0.,10,22HSUN ELEVATION ANGLE = ,0.,22)	00024040
CALL NUMBER(999.,999.,,10,SUNELV,0.,0)	00024050
CALL SYMBOL(0.,2.8.,,10,16HA/C PITCH ATT = ,0.,16)	00024060
CALL NUMBER(999.,999.,,10,THAC,0.,1)	00024070
CALL SYMBOL(0.,2.6.,,10,15HA/C ROLL ATT = ,0.,15)	00024080
CALL NUMBER(999.,999.,,10,PHAC,0.,1)	00024090
IF(((JCONT-1)/3)+3.EQ.(JCONT-1)) CALL SYMBOL(3.4,3.4,	00024100
1 0.14,19HSUN GLINT SIGNATURE,0.,19)	00024110
40 RETURN	00024120
END	00024130

G LEVEL	21	GRID	DATE = 78102	10/47/48
	SUBROUTINE GRID(XORG,YORG,DELVLS,DELHLS,NVSPS,NHSPS)			00024140
	DYMAX=NHSPS+DELHLS			00024150
	DXMAX=NVSPS+DELVLS			00024160
	XMAX=XORG+DXMAX			00024170
	YMAX=YORG+DYMAX			00024180
	NVLS=NVSPS+1			00024190
	NHLS=NHSPS+1			00024200
	XPRES=XORG			00024210
	YPRES=YORG			00024220
	DO 10 I=1,NVLS			00024230
	CALL PLOT(XPRES,YPRES,3)			00024240
	FACT=(-1)**(I+1)			00024250
	YPRES=YPRES+FACT*DXMAX			00024260
	CALL PLOT(XPRES,YPRES,2)			00024270
	XPRES=XPRES+DELVLS			00024280
	CALL PLOT(XPRES,YPRES,3)			00024290
10	CONTINUE			00024300
	XPRES=XORG			00024310
	YPRES=YORG			00024320
	DO 20 I=1,NHLS			00024330
	CALL PLOT(XPRES,YPRES,3)			00024340
	FACT=(-1)**(I+1)			00024350
	XPRES=XPRES+FACT*DYMAX			00024360
	CALL PLOT(XPRES,YPRES,2)			00024370
20	YPRES=YPRES+DELHLS			00024380
	RETURN			00024390
	END			00024400