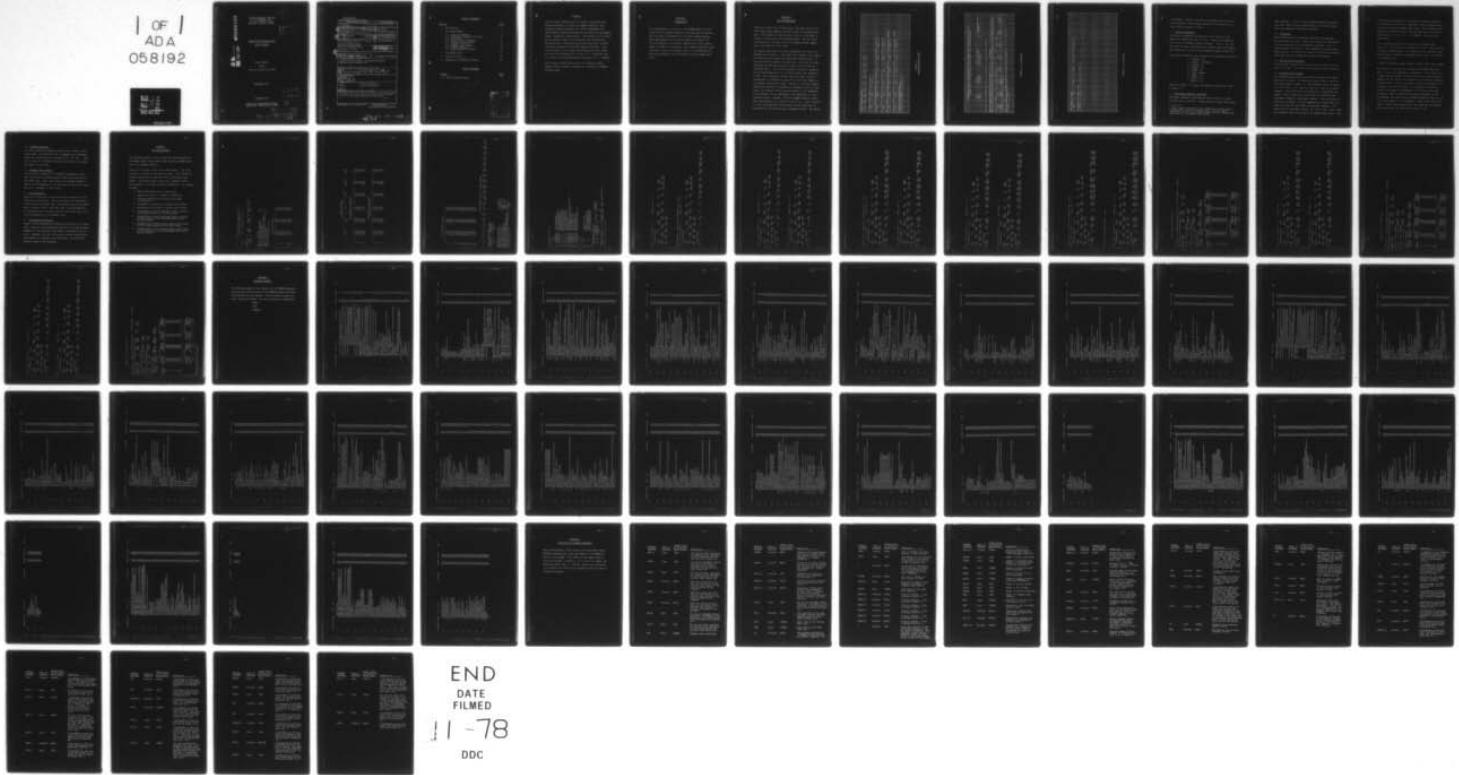


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CAYWOOD-SCHILLER DIVISION  
A. T. KEARNEY, INC.  
100 South Wacker Drive  
Chicago, Illinois 60606

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LEVEL D

DYNAMIC SAM ENDGAME MODEL

USER'S MANUAL

FINAL REPORT

Under

Contract N62269-76-C-0386

SEPTEMBER 1977

Prepared for

NAVAL AIR DEVELOPMENT CENTER  
WARMINSTER, PENNSYLVANIA 18974

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## TABLE OF CONTENTS

<u>Section</u>	<u>Page</u>
1. Introduction . . . . .	1
2. Data Requirements . . . . .	2
2.1 Control Parameters . . . . .	6
2.2 Vulnerable Component Locations . . . . .	6
2.3 Ellipsoids . . . . .	7
2.4 Glitter Point Locations . . . . .	7
2.5 Vulnerable Area Tables . . . . .	7
2.6 Fragment Densities . . . . .	9
2.7 Fragment Mass Classes . . . . .	9
2.8 Fuze Parameters . . . . .	9
2.9 Miscellaneous Constants . . . . .	9
3. New DSAMAM Output . . . . .	10
4. Program Listing . . . . .	25
5. Definition of FORTRAN Variables . . . . .	56

## LIST OF FIGURES

<u>Number</u>	<u>Page</u>
2-1 DSAMAM Endgame Inputs . . . . .	3

VII - A058 050  
CH-200 on card

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## FOREWORD

The activities described in this report in modifying the Dynamic SAM Model to include an endgame capability were performed during the period July 1976 to September 1977 under Contract N62269-76-C-0386 for the Naval Air Development Center, Warminster, Pennsylvania. The purpose of this effort was to develop a means for evaluating kill probabilities associated with trajectories and missile/aircraft intercept conditions generated by the NADC Dynamic SAM Model. This is the second of two volumes constituting the final report. The work was performed by R. H. Rose, M. A. Dloogatch, and D. S. Kluk of the Caywood-Schiller Division of A. T. Kearney.

This volume includes instructions for preparing inputs, sample outputs, program listings and a glossary of FORTRAN variable names.

SECTION 1  
INTRODUCTION

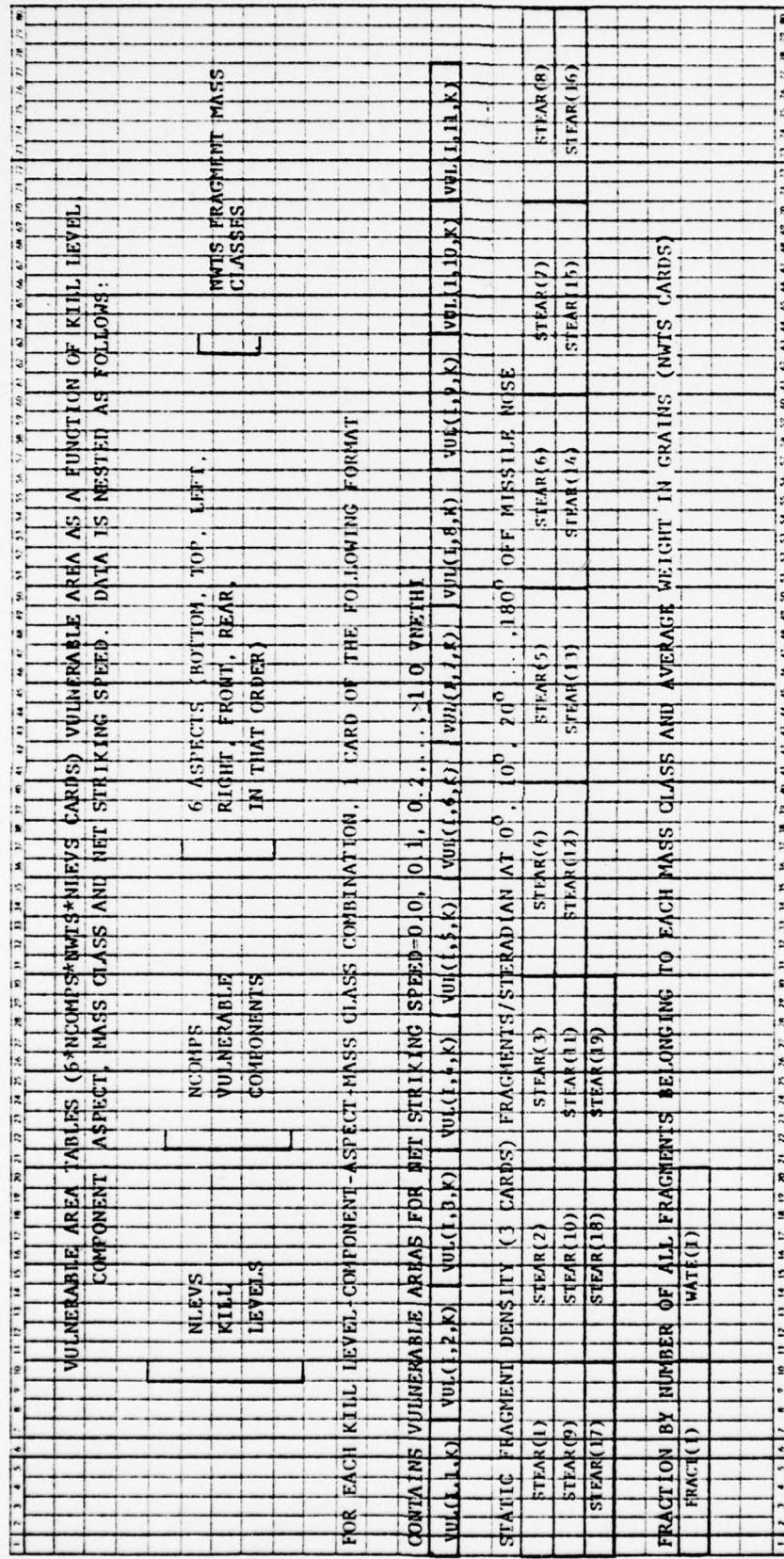
As a prerequisite to the use of the DSAMAM model with the addition of an endgame capability the user must be familiar with the input requirements for the basic DSAMAM program. These requirements are covered in detail in Volume II of the Final Report on the Dynamic SAM-Aircraft Study, Report Number C9-2336/120, Autonetics, North American Rockwell Corp. This User's Guide will confine itself to the additional operating requirements which accompany the new endgame capability.

SECTION 2  
DATA REQUIREMENTS

There are twelve sets of input cards, each made up of one or more cards, which comprise the new inputs to the DSAMAM program. The format for these inputs is shown on the forms in Figure 2-1 and each set is discussed below. The new inputs are inserted in the data for the old DSAMAM program immediately following the title cards.

All inputs other than the control parameters are read in as FORTRAN real variables. This means that normally they should be entered with an explicitly designated decimal point. If they are entered without a decimal point, they must be right-justified and the implied decimal point will be after the rightmost digit. All of the Cartesian coordinates requested on the input forms are to be in the aircraft body coordinate system. This coordinate system, under the name "stability coordinates," is defined on page 3-22 of the Final Report on the Dynamic SAM-Aircraft Study. Because in the basic DSAMAM program the aircraft is essentially considered to be a single point, the location of the origin relative to the components of the aircraft is not specified in the definition of the coordinate system. However, since the endgame addition represents the aircraft in a variety of ways (e.g., a set of points or a set of ellipsoids) the location of the origin must be specified for the aircraft body coordinate system. The choice

FIGURE 2-1  
DSAMAM ENDGAME INPUTS



**FIGURE 2-1** (continued)

**FIGURE 2-1** (continued)

is arbitrary; however, the user will probably wish the point to correspond to some natural feature of the aircraft (e.g., nose, center of gravity, pilot).

### 2.1 Control Parameters

The control parameters are read in on the first card and define the number of cards to follow. These nine parameters are read in as FORTRAN integer variables. That is, the numbers must be right-justified and no decimal points may appear. The definitions of each of the parameters appear in Section 5.

The control parameters are limited by the following constraints:

$$\begin{aligned}1 &\leq \text{NCOMPS} \leq 25^* \\0 &\leq \text{NDUBLY} \leq [\text{NCOMPS}/2] \\0 &\leq \text{NSHLDs} \leq 25 \\0 &\leq \text{NBLAST} \leq 25 \\0 &\leq \text{NDHNK} \leq 25 \\0 &\leq \text{NDHK} \leq \text{NDHNK} \\1 &\leq \text{NGLIT} \leq 25 \\1 &\leq \text{NLEVS} \leq 4^* \\1 &\leq \text{NWTS} \leq 10^*\end{aligned}$$

where the symbol [ ] is read "the greatest integer less than or equal to."

### 2.2 Vulnerable Component Locations

Vulnerable component locations are read in, one to a card, with the units to be feet. There must be at least one vulner-

---

\* These limits are based on the assumption of being able to utilize 300K (octal) of computer core. If less core is available, one or more of these numbers must be reduced (see discussion of vulnerable area inputs).

able component. If the control parameter NDUBLY is greater than zero, then the first NDUBLY pairs of vulnerable components are assumed to be doubly vulnerable.

#### 2.3 Ellipsoids

There are four sets of ellipsoid centroids and semi-axes. The units are to be feet. Each card contains the coordinates of one centroid and the corresponding semi-axes. Any of these sets may be omitted if the corresponding control parameter is zero. If there are no direct-hit-no-kill-ellipsoids, then there can be no direct-hit-with-kill-ellipsoids.

#### 2.4 Glitter Point Locations

Glitter point locations are read in, one to a card, with the units to be feet. At least one glitter point must be present.

#### 2.5 Vulnerable Area Tables

The vulnerable area tables are the most extensive and complicated set of inputs. The nesting of these cards is shown on the second page of Figure 2-1. The outermost level of nesting is by kill level. All cards for each kill level are grouped together. The second level of nesting is by vulnerable component. Within a particular kill level all cards for each vulnerable component are grouped together. The third level of nesting is by aspect. For any component at a particular kill level the cards for each aspect are grouped together. The aspects are in the order: bottom, top, left, right, front, rear. The innermost level of nesting is by fragment mass class. Thus,

for a particular aspect of a particular vulnerable component at a particular kill level, there must be one vulnerable area input card for each fragment mass class. Each individual card contains 11 vulnerable areas, one for each value of striking speed from 0.0, 0.1, 0.2, ...,  $\geq 1.0 \cdot VNETHI$ . The units are square feet.

The vulnerable area data is read into the FORTRAN array VUL(I,J,K). The subscript I denotes kill level. The subscript J denotes striking speed. The subscript K is a composite combining aspect, component and fragment mass class. The formula for computing K is:

$$K = NWTS \cdot (6 \cdot \text{component number} + \text{aspect number} - 7) + \text{mass class number}.$$

The array VUL requires core storage of  $66 \cdot NCOMPS \cdot NWTS \cdot NLEVS$  words. Thus, to accommodate 25 components, 4 kill levels and 10 mass classes requires 66,000 words for VUL alone and thus demands the full 98,304 word capacity of the CDC 6600 computer system at NADC. Since it requires a special arrangement to be allowed to use this much core, it is normally desirable to restrict the number of components, kill levels and/or mass classes in order to reduce the storage requirements of VUL. For example, the listing of the DSAMAM program in Section 4 is for a case limited to 12 components, 1 mass class, and 2 kill levels. Thus, VUL requires only 1,584 words and the program can be run using less than 40,000 words of core.

## 2.6 Fragment Densities

The static explosion fragment density data is read in from three cards. The densities are in fragments per steradian. There are values read in for angles of  $0^\circ$ ,  $10^\circ$ ,  $20^\circ$ , ...,  $180^\circ$  for a total of 19 entries (8 on each of the first two cards and three on the third).

## 2.7 Fragment Mass Classes

The fraction by number of all fragments belonging to each mass class and the average weight of each class are read in from NWTS cards. Each card contains the average weight in grains of the fragments in one class and the fraction by number of all fragments in that class.

## 2.8 Fuze Parameters

There is one card used to read in various constants for the radar and contact fuzes. This card contains the half-angle of the cone of the radar fuze, the maximum range of the radar fuze, the delay time of the radar fuze, the dud probability for the radar fuze, the delay time of the contact fuze, and the dud probability of the contact fuze.

## 2.9 Miscellaneous Constants

Finally, four miscellaneous constants are read in on one data card. These are the normalizing value for the striking speed argument in the vulnerable area tables, the range at which a miss is assumed given that the point of closest approach has been passed, the fragment drag coefficient, and the static emission speed of the fragments.

SECTION 3  
NEW DSAMAM OUTPUT

The following pages of this section show those portions of the DSAMAM output which differ from the basic DSAMAM output due to the endgame addition.

There are two types of new output which appear. The first is simply a listing of the new input data. This listing is complete except for the omission of the vulnerable area tables. The second output type is the "summary of kill" which appears at the time of missile detonation. The summary includes:

1. Time of detonation due to radar fuze.
2. Range from missile to target at detonation.
3. Inertial coordinates of missile and target at detonation.
4. Coordinates of missile in aircraft body system.
5. Coordinates of aircraft in missile body system.
6. Probabilities of each component having survived this single missile at all kill levels.
7. Probabilities of each component having survived all missiles up to and including this one at all kill levels.
8. Probabilities of the aircraft having been killed by this single missile at all kill levels.
9. Probabilities of the aircraft having been killed by all missiles up to and including this one at all kill levels.

\* FIVE CASES  
DEMONSTRATE AIR CRAFT TAI GENEERATION  
AND NEW MISSILE GUIDANCE MODE

SI FLIGHT THIS IS PEGUAG TITLE-SAMPLE FOR

INPUT DATA  
TSTART = 0.00    TFINAL = 60.00    TPILOT = .0625    TSTOP = 2.00  
LATOR = 2    LAC = 1    LPOL = 0    LPOL = 1.00  
LATR = 0    LACDIT = 1    LFDIT = 0    LFDIT = 1  
IPCE = 1

NUMBER OF COMPONENTS  
NUMBER OF CONCENTRIC ELLIPTICAL PAIRS  
NUMBER OF SATELLITE ELLIPTICS  
NUMBER OF DIRECT-HIT-OR-KILL ELLIPTICS  
NUMBER OF DIRECT-HIT-WITH-KILL ELLIPTICS  
NUMBER OF QUADRIC SURFACES  
NUMBER OF MILL LEVELS  
NUMBER OF PEGANTH REIGAT CLASSES

	X	Y	Z
1	-6.70	0.00	4.70
2	-34.10	0.00	1.80
3	-29.30	0.00	1.40
4	-17.90	1.90	3.50
5	-17.90	-1.90	3.50
6	-23.50	1.90	3.60
7	-23.50	-1.90	3.60
8	-28.20	0.00	3.10
9	-23.20	6.60	5.60
10	-23.20	-4.60	5.60
11	-35.30	0.00	3.70
12	-32.30	0.00	3.70

BLAST ELLIPSOIDS

CENTROIDS AND SEMI-AXES

	X	Y	Z	A	B	C
1	-23.46	6.00	5.09	35.64	40.09	35.64

DIRECT-HIT-NO-KILL ELLIPSOIDS

CENTROIDS AND SEMI-AXES

	X	Y	Z	A	B	C
1	-23.10	6.00	5.70	23.10	3.00	4.00
2	-6.1.6.0	0.00	11.10	3.60	*.20	3.80
3	-4.4.4.0	0.00	15.30	2.60	*.13	2.80
4	-25.5.0	6.80	5.40	4.70	*.70	*.50
5	-25.5.0	-6.80	5.40	4.70	*.70	*.50
6	-29.7.0	15.30	5.40	4.00	*.00	*.30
7	-29.7.0	-15.30	5.40	4.00	*.00	*.30
8	-4.0.8.0	0.00	3.50	4.30	*.31	*.20

DIRECT-HIT-WITH-KILL ELLIPSOIDS

CENTROIDS AND SEMI-AXES

	X	Y	Z	A	B	C
1	-23.10	6.00	5.00	23.10	3.00	4.00
2	-4.1.6.0	0.00	11.10	3.60	*.20	3.80
3	-6.4.4.0	0.00	15.30	2.60	*.13	2.80
4	-25.5.0	6.80	5.40	4.70	*.70	*.50
5	-25.5.0	-6.80	5.40	4.70	*.70	*.50
6	-29.7.0	15.30	5.40	4.00	*.00	*.30
7	-29.7.0	-15.30	5.40	4.00	*.00	*.30
8	-4.0.8.0	0.00	3.50	4.30	*.31	*.20

GAUJAE POINT LOCATIONS

	X	Y	Z
1	-4.33	1.70	2.50
2	-3.33	-1.70	2.50
3	-4.00	0.00	6.90
4	-31.50	19.60	7.60
5	-31.50	-19.60	7.60
6	6.00	2.37	7.20
7	-15.90	-2.33	7.20
8	-30.10	2.20	7.60
9	-20.10	-2.20	7.60
10	-42.80	0.00	9.70
11	-44.20	0.00	7.50
12	-43.40	0.00	18.26
13	-40.50	6.63	6.20
14	-40.50	-6.63	6.20
15	-36.36	7.33	5.50
16	-36.36	-2.33	5.30
17	-43.70	2.00	5.30
18	-43.70	-2.00	5.30
19	-3.20	0.00	6.50
20	-10.60	0.00	6.60
21	-7.63	0.00	5.00
22	-19.60	0.00	1.60
23	-32.80	0.00	2.10

STATIC FRAGMENT DENSITY BY 10 DEGREE INTERVALS

	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180
AVERAGE WEIGHT IN GRAMS	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.

FRAGMENT DISTRIBUTION BY WEIGHT  
AVERAGE WEIGHT FRACTION OF  
IN GRAMS ALL FRAGMENTS  
145.0 1.00

HOLD-ANGLE OF GAUJAE FUZE  
MAXIMUM RANGE OF GAUJAE FUZE  
DELAY TIME OF GAUJAE FUZE  
TIME FOR IGNITION OF GAUJAE FUZE  
DELAY TIME OF CONTACT FUZE  
TIME FOR IGNITION OF CONTACT FUZE  
ROPE LENGTH SUPPLIED FOR VULNERABLE AREA TABLE  
DISTANCE ASSUMED AFTER FUZE TO FUZE WHEN RANGE = 800. FEET  
FOR EACH FUZE FUZE LENGTH = 1920. FEET  
FUZE LENGTH = 1920. FEET

\*\*\*\*\*MISSILE TYPE 2\*\*\*\*\*

LIST OF EXIT

J	TEST(J)	TEST(J)
1	1. 000000E+03	NEVER BEGIN MANEUVER
24	2. 000000E+00	TIME BETWEEN MISSILES (SEC.)
25	1. 000000E+03	NEVER FIRE SECOND SALVO
41	1. 000000E+05	MAXIMUM LAUNCH RANGE (FT)
42	0.	MINIMUM LAUNCH RANGE (FT)
43	5. 000000E+04	MAX ALTITUDE FOR LAUNCH (FT)
44	0.	MIN TARGET ALTITUDE (FT)
45	9. 000000E+01	MAX LAUNCH ELEVATION (DEG.)
46	0.	MIN TARGET ELEVATION (DEG.)
52	1. 200000E+04	RANGE FROM MISSILE TO A/C BEGIN HOMING (FT)
54	1. 000000E+04	FRONT VULNERABLE AREA (50 FT)
55	0. 000000E+01	REAR VUL. AREA (50 FT)
56	6. 000000E+02	SIDE VUL. AREA (50 FT)
57	1. 000000E+02	BOTTOM VUL. AREA (50 FT)
58	6. 000000E+03	TRACKING ERROR (RAD)
59	1. 000000E-03	ERROR FOR HOMING GUIDANCE (FT)
60	2. 000000E+01	

LIST OF FIXES

J	TEST(J)	TEST(J)
11	2	= MISSILES IN SALVO
12	2	= SALVO
13	1	= DO NOT DETECT OR ACQUIRE
26	20	= MISSILES ALLOWED IN AIR
24	0	= SUPPRESS ANGULAR RATE TEST
27	0	= SUPPRESS SPP & CLOSING TEST

INITIAL ATC/CAFI STATE VECTOR

POSITION	-2.500000E+04	0.	5.000000E+03
	2.542510E+04	3.141592E+00	1.973456E-01
VELOCITY	5.000000E+02	0.	0.
	5.000000E+02	0.	0.

AT ATTITUDE OF SAM SITE 0. FT. STRUCTURE LIMIT 2.0 G'S THROTTLE SETTING .30  
NUMBER OF TIME LAGS 1. TIME LAGS IN SEC. .100

\*\*\*\*\*-4 IS A/C AFTER WAKE\*\*\*\*\*

TYPE= 0.000 A/C FLAGS- 1 1 1 SAM SITE FLAGS- 1 1 1

AIRCRAFT VARIATECS

	X	Y	Z	RHAG	VX	VY	VZ	VMAG	
-25 J00.0	0.	0.	5000.0	25495.	0.	0.	0.	500.0	
HACH	ALPHA	GFORCE	THUST	THROT	LIFT	DRAG	CL	CD	CLM
.4f	0.01	0.00	0.	0.	0.	0.	0.000	0.000	0.000
ACOM	AOUTA	AOUTD	AOUT	VDDT					
0.0	3.0	0.0	0.0	0.0					

\*\*\*MISSION NO. 1 REBURN NO. 1 IS LAUNCHED AT TIME = 0.000

TYPE= 0.000 A/C FLAGS- 1 1 1 SAM SITE FLAGS- 0 0 0

AIRCRAFT VARIATECS

	X	Y	Z	RHAG	VX	VY	VZ	VMAG						
-0.000.0	0.	0.	5000.0	25495.	0.	0.	0.	500.0						
HACH	ALPHA	GFORCE	THUST	THROT	LIFT	DRAG	CL	CD	CLM					
.4f	0.00	0.00	0.	0.	0.	0.	0.000	0.000	0.000					
ACOM	AOUTA	AOUTD	AOUT	VDDT										
0.0	0.0	3.0	0.0	0.0										
NO. X Y Z	0.	0.	-9.1	VX	VY	VZ	VMAG	MACP	PREL	KDST	LAMDAE	LAMEDAA	UELE	UFLA
1	0.	0.	-9.1	0.	0.	0.	0.00	0.00	0.00	0.0	0.00	0.000	0.000	0.000

TIME = 2.000 ATCRAFT FLAGS - 2 1 1 SAM SITE FLAGS - 5 1 0

ATCRAFT VARIABLES

	RX	RY	RZ	RHAG	VX	VY	VZ	VMAG
-2 1999.9	0.0	5000.0	24515.	5.00.	0.	0.0	0.0	500.1
MACH	ALPHA	GFORCE	THRUST	LIFT	DRAG	CL	CD	CLMX
.46	5.30	1.00	5493.	*30	4.0004.	5429.	.294	.040 .970
ACOM	ACUTA	AUTOT	AOUT	VOUT				
0.0	3.0	0.0	0.0	0.0				

MISSILE OUTPUT

	NO.	X	Y	Z	VX	VY	VZ	VMAG
1	-769.	-6.	269.	-797.0	-0.0	237.2	631.5	0.72 23710. 1318.3

TIME = 4.000 ATCRAFT FLAGS - 2 1 1 SAM SITE FLAGS - 5 1 0

ATCRAFT VARIABLES

	RX	RY	RZ	RHAG	VX	VY	VZ	VMAG
-2 2999.7	0.0	5000.0	23537.	5.00.	0.	0.0	0.0	500.1
MACH	ALPHA	GFORCE	THRUST	LIFT	DRAG	CL	CD	CLMX
.46	5.30	1.00	5494.	*30	4.0004.	5429.	.294	.040 .970
ACOM	ACUTA	AUTOT	AOUT	VOUT				
0.0	3.0	0.0	0.0	0.0				

MISSILE OUTPUT

	NO.	X	Y	Z	VX	VY	VZ	VMAG
1	-3258.	-0.	917.	-1704.9	-0.0	416.2	1755.0	1.55 20159. 2243.7

\*\*\*\*\*MISSION NO. 2 TODAY TO. 21 IS LAUNCHED AT TIME = 5.000

TIME= 6.300 A/C FLAGS= 2 1 1 SAM SITE FLAG= 4 1 0

AIRCRAFT VARIABLES

SIPFLT

	VX	RY	RZ	RHAG	VX	VY	VZ	VMAG
	-2499.6	0.0	5000.0	2306.0.	.500.	0.	0.0	500.2
MACH	ALPHA	GFORCE	THRUST	THROTL	LIFT	Drag	CL	CD
.66	5.30	1.00	5494.	.30	40004.	5430.	.294	.040
ACM	AOUA	AOUT	AOUT	VDOT				
0.0	0.0	0.0	0.0	0.0				

MISSILE OUTPUT

	X	Y	Z	VX	VY	VZ	VMAG	MACH	PREL	K001	LAMDAE	LAMDAA	DELE	DELA
1	-5.103.	-0.	1.351.	-1.062.1	-.0	4.26.7	1929.9	1.74	17775.	2419.2	0.00	0.00	0.000	0.000
2	0.	0.	0.	-.5.0	-.0	4.4	10.0	0.00	23048.	0.0	0.00	0.00	0.000	0.000

TIME= 6.300 A/C FLAGS= 2 1 1 SAM SITE FLAG= 5 2 0

AIRCRAFT VARIABLES

SIPFLT

	VX	RY	RZ	RHAG	VX	VY	VZ	VMAG
	-21995.4	0.0	5000.0	22500.	.500.	0.	0.0	500.2
MACH	ALPHA	GFORCE	THRUST	THROTL	LIFT	Drag	CL	CD
.66	5.30	1.00	5494.	.30	40004.	5430.	.294	.040
ACM	AOUA	AOUT	AOUT	VDOT				
0.0	0.0	0.0	0.0	0.0				

MISSILE OUTPUT

	X	Y	Z	VX	VY	VZ	VMAG	MACH	RRFL	K001	LAMDAE	LAMDAA	DELE	DELA
1	-6.956.	-1.	1.764.	-1.907.3	-.0	399.2	1946.7	1.75	15349.	2433.6	0.00	0.00	J.000	0.000
2	-1.92.	-0.	.75.	-3.76.0	-.0	146.3	400.4	.34	22366.	884.5	0.00	0.00	J.000	0.000

TIRE = 8.000 A/C FLAGS= 2 1 1

AIRCRAFT VARIABLES

SIRFLT

	RX	RY	RZ	RHAG	VX	VY	VZ	VMAG
MACH	0.0	5.000.0	21586.	5.00.	0.	0.	0.0	500.3
ACOM	ALPHA	GFORCE	THRUST	LIFT	DRAg	CL	CD	CLMX
1.0	5.5	1.00	5494.	.30	40004.	.294	.040	.970
MISSILE OUTPUT	X	Y	Z	VX	VY	VZ	VMAG	VMAG
No.	1 -1084.3	0.	2520.	-1940.0	.0	375.6	1976.0	1.78
2 -1773.	-6.	578.	-1233.4	-0.0	554.6	1283.3	1.13	2459.7
						19726.	1769.0	0.00
							0.00	0.000

TIRE = 10.000 A/C FLAGS= 2 1 1

AIRCRAFT VARIABLES

SIRFLT

	RX	RY	RZ	RHAG	VX	VY	VZ	VMAG
MACH	0.0	5.000.0	20614.	5.00.	0.	0.	0.0	500.3
ACOM	ALPHA	GFORCE	THRUST	LIFT	DRAg	CL	CD	CLMX
0.0	0.0	0.0	0.0	0.0	0.0	.294	.040	.970
MISSILE OUTPUT	X	Y	Z	VX	VY	VZ	VMAG	VMAG
No.	1 -1476.1	0.	3320.	-1952.5	.0	453.9	2034.6	1.82
2 -5968.	-6.	1451.	-1870.6	-0.0	464.0	1927.3	1.73	2475.0
						15346.	2416.0	0.00
							0.00	0.000

\*\*\*\*\*MISSILE NO. 3 BODY no. 3 IS LAUNCHED AT TIME = 19.000

TIME= 19.000 A/C FLAGS- ? 1 SAM SITE FLAGS- 4 2 0

AIRCRAFT VARIABLES

STRFLT

	X	Y	Z	PMAg	VX	VY	VZ	VMAG
MACH	ALPHA	GFORCE	THRUST	THROTL	LIFT	DRAG	CL	CD
.46	5.29	1.06	54.94.	.30	4.0004.	54.31.	.294	.040 .974
ACOM	AUTIA	AUTO	AOUT	VDO1				
0.0	0.0	0.0	0.0	0.0				

MISSILE OUTPUT

NO.	X	Y	Z	VX	VY	VZ	VMAG
1	-1.6741.	0.	33.28.	-1952.5	*.0	453.9	2604.6
2	-5.068.	-0.	1.451.	-1870.6	-.0	466.0	1927.3
3	0.	0.	0.	0.	-.0.9	4.6	10.0

MISSILE OUTPUT

NO.	X	Y	Z	VX	VY	VZ	VMAG
1	-1.6741.	0.	33.28.	-1952.5	*.0	453.9	2604.6
2	-5.068.	-0.	1.451.	-1870.6	-.0	466.0	1927.3
3	0.	0.	0.	0.	-.0.9	4.6	10.0

LAST MISSILE IN SALVO FIRED AT TIME = 16.000

TIME= 16.000 A/C FLAGS- ? 1 1 SAM SITE FLAGS- 6 3 0

AIRCRAFT VARIABLES

STRFLT

	X	Y	Z	PMAg	VX	VY	VZ	VMAG
MACH	ALPHA	GFORCE	THRUST	THROTL	LIFT	DRAG	CL	CD
.46	5.29	1.06	54.94.	.30	4.0004.	54.31.	.294	.040 .970
ACOM	AUTIA	AUTO	AOUT	VDO1				
0.0	0.0	0.0	0.0	0.0				

MISSILE OUTPUT

NO.	X	Y	Z	VX	VY	VZ	VMAG
1	-1.6525.	0.	6.5649.	-1675.1	-.0	915.9	1910.9
2	-5.052.	-0.	2.325.	-1.915.8	-.0	409.7	1957.2
3	-7.44.	-0.	2.91.	-7.83.2	-.0	271.7	829.0

## SUMMARY OF KILL OUT TO MISSILE No. 1

\*\*\*MISSILE No. 1 BURSTS OUT TO PARAP FORT AT TIME = 12.270 AND RANGE =

191.0 FEET

STATE VECTORS IN INERTIAL COORDINATES AT BURST			
X	Y	Z	V <sub>X</sub>
AIRCRAFT -1.9462.	1.	50.60.	0.0
MISSILE -1.9463.	0.	48.38.	-154.75

MISSILE IN A/C BODY COORDINATES

X	Y	Z	ELEVATION	AZIMUTH
-114.16	0.00	-152.42	-52.93	180.00

A/C IN MISSILE BODY COORDINATES

X	Y	Z	ELEVATION	AZIMUTH
56.46	-0.00	-102.50	72.61	0.00

MISSILE STATUS AT BURST

OPFG	AM1	ANY	0FT	AP
7.724E-10	-7.430E+00	-5.520E+02	-1.000E+01	1.931E+01
7.055E-01	6.190E-07	-8.771E-08	-5.151E-09	2.092E-09
-3.228E-10	8.670E+00			

## KILL LEVEL 1

COMPONENT	SINGL-SHOT SURVIVAL PROBABILITY	CUMULATIVE SURVIVAL PROBABILITY	SINGLE-SHOT SURVIVAL PROBABILITY	CUMULATIVE SURVIVAL PROBABILITY
1	.996	.996	.989	.989
2	1.000	1.000	.999	.999
3	1.000	1.000	.995	.995
4	1.000	1.000	.990	.990
5	1.000	1.000	.950	.950
F	1.000	1.000	.967	.967
7	1.000	1.000	.967	.967
6	1.000	1.000	.903	.903
3	1.000	1.000	.946	.946
19	1.000	1.000	.966	.966
11	1.000	1.000	.996	.996
12	1.000	1.000	.994	.994

## KILL LEVEL 2

COMPONENT	SINGL-SHOT SURVIVAL PROBABILITY	CUMULATIVE SURVIVAL PROBABILITY	SINGLE-SHOT SURVIVAL PROBABILITY	CUMULATIVE SURVIVAL PROBABILITY
-----------	---------------------------------	---------------------------------	----------------------------------	---------------------------------

1	.996	.996	.989	.989
2	1.000	1.000	.999	.999
3	1.000	1.000	.995	.995
4	1.000	1.000	.990	.990
5	1.000	1.000	.950	.950
F	1.000	1.000	.967	.967
7	1.000	1.000	.967	.967
6	1.000	1.000	.903	.903
3	1.000	1.000	.946	.946
19	1.000	1.000	.966	.966
11	1.000	1.000	.996	.996
12	1.000	1.000	.994	.994

TIME = 16.000 A/C FLAGS- 2 1 1 SAM SITE FLAGS- 6 3 0

AIRCRAFT VARIABLES

	RX	RY	RZ	RHAB	VX	VY	VZ	VMAG
MACH	-1.694e+0	0.0	5.000	18679.	500.	0.	0.0	500.4
ALPHA	5.29	1.00	5.49e+	THROTL	LIFT	DRAg	CL	CD
.46				*.30	*.0004.	5.432.	.294	.040
ACOM	AUDIA	AUDIO	AUDI	VNOI				
0.0	J.J	0.0	0.0	0.0				
MISSILE OUTPUT								
NO.	X	Y	Z	VX	VY	VMAG	MACH	RRLL
2	-1.2997.	0.	3199.	-1922.8	*.0	1.98e+.4	1.80	2452.2
3	-3265.	-0.	1059.	-1679.7	-.0	4.91e+.4	1750.1	1.55
							15308.	2233.1
							0.00	0.000

TIME = 16.000 A/C FLAGS- 2 1 1 SAM SITE FLAGS- 6 3 0

AIRCRAFT VARIABLES

	RX	RY	RZ	RHAB	VX	VY	VZ	VMAG
MACH	-1.694e+0	0.0	5.000	17716.	501.	0.	0.0	500.5
ALPHA	5.29	1.00	5.49e+	THROTL	LIFT	DRAg	CL	CD
.46				*.30	*.0004.	5.432.	.294	.040
ACOM	AUDIA	AUDIO	AUDI	VNOI				
0.0	0.0	0.0	0.0	0.0				
MISSILE OUTPUT								
NO.	X	Y	Z	VX	VY	VMAG	MACH	RRLL
2	-1.641e-	0.	4539.	-1648.9	-.0	1.905e+.7	1.75	2257.6
3	-6998.	-0.	2071.	-1876.8	-.0	483.6	1938.1	1.75
							10523.	2418.0
							0.00	0.000
							0.30	0.000

## SUMMARY OF KILL DUE TO MISSILE NO. 2

\*\*\*\*MISSILE NO. 2 EFFECTS DUE TO PACIFIC FIGHT AT TIME = 16.750 AND RANGE =

167.6 FEET

## STATE VECTORS IN INERTIAL COORDINATES AT BURST

	X	Y	Z	VX	vy	VZ
AIRCRAFT	-16831.	0.	5000.	500.5	6.0	0.0
MISSILE	-16936.	0.	4845.	-1495.1	-0.0	1106.4

## MISSILE IN A/C BODY COORDINATES

	X	Y	Z	ELEVATION	ATTITUDE
A/C	-120.46	.00	-144.10	-50.11	180.00
MISSILE	23	.00	-179.52	72.91	.00

## MISSILE STATUS AT BURST

C-MFG	AMT	ANY	DET	AN
3.753E-10	-3.790E+02	-5.771E+02	-1.000E+01	1.519E+01
6.103E-01	5.704E+07	-8.150E-08	-5.615E-09	1.975E-09
-3.00E-15	8.846E+02			

CURRENT NUMBER	KILL LEVEL 1	SINGLE-SHOT SURVIVAL PROBABILITY	KILL LEVEL 2	AIRCRAFT
1	.992	.988	.974	.963
2	1.000	1.000	.997	.996
3	1.000	1.000	.989	.949
4	1.000	1.000	.981	.837
5	1.000	1.000	.879	.835
6	1.000	1.000	.917	.687
7	1.000	1.000	.915	.685
8	1.000	1.000	.752	.679
9	1.000	1.000	.914	.683
10	1.000	1.000	.914	.883
11	1.000	1.000	.987	.983
12	1.000	1.000	.981	.975

CURRENT NUMBER	KILL LEVEL 1	SINGLE-SHOT SURVIVAL PROBABILITY	KILL LEVEL 2	AIRCRAFT
1	.992	.988	.974	.963
2	1.000	1.000	.997	.996
3	1.000	1.000	.989	.949
4	1.000	1.000	.981	.837
5	1.000	1.000	.879	.835
6	1.000	1.000	.917	.687
7	1.000	1.000	.915	.685
8	1.000	1.000	.752	.679
9	1.000	1.000	.914	.683
10	1.000	1.000	.914	.883
11	1.000	1.000	.987	.983
12	1.000	1.000	.981	.975

CURRENT NUMBER	KILL LEVEL 1	SINGLE-SHOT SURVIVAL PROBABILITY	KILL LEVEL 2	AIRCRAFT
1	.992	.988	.974	.963
2	1.000	1.000	.997	.996
3	1.000	1.000	.989	.949
4	1.000	1.000	.981	.837
5	1.000	1.000	.879	.835
6	1.000	1.000	.917	.687
7	1.000	1.000	.915	.685
8	1.000	1.000	.752	.679
9	1.000	1.000	.914	.683
10	1.000	1.000	.914	.883
11	1.000	1.000	.987	.983
12	1.000	1.000	.981	.975

TIME = 18.000 A/C FLAGS- 2 1 1 SAM SITE FLAGS- 6 3 0

AIRCRAFT VARIABLES

	VX	RY	RZ	PHAG	VX	VY	VZ	VMAG
MACH	ALPHA	GEODE	THRUST	LIFT	DRAG	CL	CD	S00.6
.46	5.29	1.00	5494.	.30	.0005.	.0452.	.294	.040 .970
ACOM	AUDIA	AUDIB	AUDI	VOOT				
0.0	0.0	0.0	0.0	0.0				

MISSILE OUTPUT

NO.	X	Y	Z	VX	VY	VZ	VMAG	MACH	RRFL	ROOT	LAMDAAS	ULLE	DFLA
2 -1857.	0.	30.21.	-1894.7	0.	519.1	1964.5	1.78	5675.	2425.9	716.02	*00	-1.950	-0.00

TIME = 20.000 A/C FLAGS- 2 1 1 SAM SITE FLAGS- 6 3 0

AIRCRAFT VARIABLES

	VX	RY	RZ	PHAG	VX	VY	VZ	VMAG
MACH	ALPHA	GEODE	THRUST	LIFT	DRAG	CL	CD	S00.6
.46	5.29	1.00	5494.	.30	.0005.	.0453.	.294	.040 .970
ACOM	AUDIA	AUDIB	AUDI	VOOT				
0.0	0.0	0.0	0.0	0.0				

MISSILE OUTPUT

NO.	X	Y	Z	VX	VY	VZ	VMAG	MACH	RRFL	ROOT	LAMDAAS	ULLE	DFLA
3 -14316.	0.	4469.	-1597.6	0.	932.7	1681.1	1.72	960.	2234.1	7014.00	*0C-10.000	-0.00	

## SUMMARY OF KILL DUE TO MISSILE NO. 3

\*\*\*MISSILE NO. 3 SHOT DOWN TO RALAR FUSE AT TIME = 21.40? AND RANGE = 269.2 FEET

STATE VECTORS IN INERTIAL COORDINATES AT PBURST  
 AIRCRAFT -14792. X 0. Y 0. Z 4839.  
 MISSILE -14976. X 0. Y 0. Z 4839.

## MISSILE IN A/C BODY COORDINATES

X Y Z ELEVATION AZIMUTH  
 -147.66 .30 -148.14 -45.09 180.00

A/C IN MISSILE BODY COORDINATES  
 X Y Z ELEVATION AZIMUTH  
 -2.83 -.80 -203.43 73.36 .00

## MISSILE STATUS AT PBURST

COMPONENT	AM1	ANY	DEL	AB
0.958E-11	-7.0569E+02	-6.408E+02	-1.090E+01	1.705E+01
6.832E-01	6.034E-01	-6.365E-08	-4.536E-09	1.523E-09
-2.00E-16	6.036E-02			

KILL LEVEL 1 COMPONENT NUMBER	CUMULATIVE SURVIVAL PROBABILITY	KILL LEVEL 1 SINGLE-SHOT SURVIVAL PROBABILITY	KILL LEVEL 2 SINGLE-SHOT SURVIVAL PROBABILITY	KILL LEVEL 2 CUMULATIVE SURVIVAL PROBABILITY
1	.989	.977	.966	.951
2	1.000	1.000	.990	.992
3	1.000	1.000	.937	.711
4	1.000	1.000	.876	.699
5	1.000	1.000	.876	.697
6	1.000	1.000	.861	.781
7	1.000	1.000	.847	.779
8	1.000	1.000	.847	.736
9	1.000	1.000	.801	.778
10	1.000	1.000	.881	.778
11	1.000	1.000	.379	.962
12	1.000	1.000	.970	.965

KILL LEVEL 1 AIRCRAFT KILL PROBABILITY	AIRCRAFT KILL PROBABILITY	KILL LEVEL 2 AIRCRAFT KILL PROBABILITY	AIRCRAFT KILL PROBABILITY
.013	.023	.014	.953

SALVO NO. 1 COMPLETE AT TIME = 26.636  
 HEAT SALVO BEGINS AT TIME = 1020.436

SECTION 4  
PROGRAM LISTINGS

The following pages of this section are the FORTRAN program listings for those portions of the DSAMAM program which have been modified for the endgame. This includes the main program, subroutine MISSILE, and the following new subroutines:

- . SPRAY
- . STAB
- . CONFUZE

FTN 4.6\*4.20      77.0306 • 16.10.46      PAGE 1

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1 PROGRAM CTRANSITIN01,CDIOUT=137,TAPE1,TAPE2,TAPE3,TAPET,16,
1                   TAPE5=14001,TAPE6=OUTPUT)
1 COMMON/L11IVE/L1M,LM,ISTAK,IOTAK,IOTAT,EXTR(60),TEXTR(30),TITLE(18),
1                   SEG,GP1,RA0
1 COMMON/FLAGS/BLUL,FLUL,JSAM,KSAM,LSAM,NMIN,NTOP,FAC,EDIT,
1                   IGUFE,IPR0,MEFAC(120),JW
1 COMMON/ATKCF/FA01VA(6),ACOM,ACOMDAGOM,AOUTA,AOUTD,AOUT,VDT,
1                   UNIVTSY,A13,D43,UNIVTSY,A13,D131,01131,MACH,OA,
2                   ALPHA,DTT,DTA,G,GECE,UNIL(43),CL,CD,CLN,X,SA,CA,
3                   THRS,THROTL,ASPAK,ACTION,IPASPK,DUMPL,BURGL,TTAU,
4                   TAUT2,AN0(2,2)
4 COMMON/ACAV/TULTE(13,21),C00(21),CNCCL2(21),PAGHM(13),ALPHA0(21)
1                   ,DAUFC(21),CLMAX(21),AREA,WEIGHT,
1 COMMON/H15SL/FN(6,20),VN(6,20),MACPH(20),QAP(20),SSA,AREAN,
1                   W(12,20),DE(12,20),AD(12,20),HU(20),RAH(20),WHD(20)
1 COMMON/ACTNC/X(3,6,11),AT(9,4,81),FURN(6,2),NPTS,INDUM(16),OP(3)
1 COMMON/EPINJ/LJ,LJX
COMMON/ZONEF/RCOME,S,HEVS,WHIS,NSH05,NDLAST,NG11,THA,ROCHA,UE,
1 PEL,T,XCM,VCP,ZCP,VENH,I
COMMON /TH0/ XG(125),YG(125),ZG(125),YSH(25),YSH(25),ZSH(25),
1 ASH(25),BSH(25),CSH(25),DSH(25),YBL(25),ZBL(25),ABL(25),
2 GBL(25),XNCCL0(20),YGBL0(20),EX(25),WY(25),ZL(25),
3 TMA(9),TMU(9)
COMMON /THREE/ HODDLY,NUHNUK,NUHNUK,NUHNUK,NUHNUK,NUHNUK,NUHNUK,
1 RMSS,CDF
25 COMMON /FOUR/ X0(4,5,2),YC(25,2),Z0(65,2),Y5(25,2),Z5(25,2),
1 ZS(25,2),WATE(10)
COMMON /FIVE/ VOL(2,11,72),STEAK(19),FRAC(110),STOM(10)
COMMON /SIX/ CPS(4,25)
COMMON /SEVEN/ TMU(9)
DIMENSION THE(19)
REAL LIFT,MACH,MACHX,MACHXX
INTEGER AFLAC(16)
EQUIVALENCE AFLAC(16),JFOL
PICAL SVST(12),VPAWH(3,20)
REAL XA1(16),XA1E(1897)
EQUIVALENCE XA1(16),XA1E(1897)
FULTER JX(13),TCOBJ(12)
REAL PARM(3,3),DRAME(6),ZMORE(4)
INTEGER TPAWH(3),JPARM(3)
LOGICAL IPU(8)
EQUIVALENCE JX(1),J1,JX(2),J2,JX(3),J3,JX(4),ACTION
DATA BLANK/1H /
DATA CO/1H/
DATA SHD/4H+END/,ACTN/0H+END/
NAMELIST/INPUT/IO1,IO2,ICORD,IGUFE,IPRINT,IAC,
1                   IPLOT,OPLOT,JACIN,IACOUT,EDIT,ISIOR,IMODE
62                   30
63                   C*****INITIALIZE CONSTANTS
64                   0.0=57.29578
65                   6=52.376
66                   P1=ATAN2(0,0,-1,0)
67                   RAD=1.7453291-02
68                   A(3)=6.
69                   C*****END OF INITIALIZATION

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PROGRAM DYNTRAN    R4.74   OPT=1          77/03/04. 14.10.44      PAGE
                                                 RTH 4.66420

55      DO 50 J=1,30                         SAM 42
      EXTR(J,30)=0.                         SAM 43
50      IF(XF(J,0)=0                         SAM 44
      C***** INITIALIZE NAMELIST FLAGS
      IPLOT=0                                SAM 45
      ISTART=0                               SAM 46
      IACIN=0                                SAM 47
      LFOR=0                                 SAM 48
      LFOUT=0                               SAM 49
      RTOUT=0                               SAM 50
      RTINT=0                               SAM 51
      IPRINT=1                             SAM 52
      TAC=1                                 SAM 53
      IGUODE=1                            SAM 54
      ICODE=1                               SAM 55
      ICODEC(1)=1                          SAM 56
      ICODEC(2)=1                          SAM 57
      SAM 58
C***** READ NAMELIST, COMMENT, AND TITLE
      100 READS, INPUT
      IF(TOTAL.LT.0.) GO TO 1000
C***** SET STORAGE FLAG IF TRAJECTORY OUTPUT IS DESIRED
      IF(ILACOUT .GT. 0) ISTORE=1
      WRITE(6,600)
600     FORMAT(1H)
      101 READ(5,500) TITLE
500     FORMAT(10A4)
      IF(TITLE(11) .NE. ' ') GO TO 103
      WRITE(6,601) TITLE
      601 FORMAT(1Y,10A4)
      GO TO 101
      103 WRITE(6,602) TITLE
      602 FORMAT(1H0,10A4)
      WRITE(6,603) ISHART, TOTAL, MM, DPPO, IGUODE, IAC, IPLOT, OPLOT, JACIN,
      1      TACOUT, EDIT, ISTORE, IMODE
603     FORMAT(1H0)INPUT DATA,IH,ISTART = F7.2,10H   TOTAL = F9.2,3X,
      1      THHM = F7.4,2X,6HCIPD = FF,2/5X,SHGUODE = F5.3X,
      1      SHAC = F5.3X,7HPL01 = F5.3X,7HPL01 = F7.2,5Y,
      3      ZHACIN = F5.3X,8H1ACOUT = F5.3X,7HISUD1 = F5.3X,
      3      ZHISTOR = F5.5X,7HIPDCE = F5.3X
      4      READ(5,200)ICOMP,MMBLTY,NBLAST,NNBK,NNLT,NNEV,S,
      4      INWFS
      200 FORMAT(9(3X,12))
      WRITE(6,500)ICOMP,MMBLTY,NBLAST,NNBK,NNLT,NNEV,S,
      1NBLTS
      300  FORMAT(//," NUMBER OF COMPONENTS",22X,"=",13/
      1," NUMBER OF OCCUPY VULNERABLE PAIRS",9X,"=",13/
      2," NUMBER OF SHIELDING ELLIPSoids",12X,"=",13/
      3," NUMBER OF BLAST ELLIPSoids",16X,"=",13/
      4," NUMBER OF DIRECT-HIT-KILL EFFICIENCIES",13X,
      5," NUMBER OF DIRECT-HIT-KILL EFFICIENCIES",13X,
      6," NUMBER OF GLITTER FONTS",16X,"=",13/
      7," NUMBER OF KILL LEVELS",21X,"=",13/
      195

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PROGRAM EYUSAM      T4/T4      OPT=1      FIN 4.6•4.20      77/08/04 • 14.10.44      PAGE      3

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      9" NUMBER OF FRAGILE WEIGHT CLASSES",9X,"="13)
C*****READ COMPONENT LOCATIONS
      READ(5,2001)(FX(K),WY(K),ZE(K),K=1,NCOMP5)
110      2001 FORMAT(5F10.0)

      WRITE(6,3001)
      *011 FORMAT(//16X,"COMPONENT LOCATIONS")
      WRITE(6,3002)(K,EX(K),DY(K),ZE(K),K=1,NCOMPS)
      *022 FORMAT(19X,"X","Y","14X","14X","2"/5X,45("=")/15,3F15.2)
C*****READ SHIELD ELLIPSOID LOCATIONS AND SEMIAXES
      READ(5,2002)(XSH(K),YSH(K),ZSH(K),ASH(K),BSH(K),CSH(K),K=1,NSHLD5)
      *022 FORMAT(6F10.0)
      WRITE(6,3003)
      *033 FORMAT("1",33X,"FRAGMENT SHIELDING ELLIPSOIDS")
      WRITE(6,3004)(K,XSH(K),YSH(K),ZSH(K),ASH(K),BSH(K),CSH(K),
      1K=1,NSHLD5)
      *006 FORMAT(37X,"CENTROIDS AND SEMI-AXES",19X,"X",
      "1A",14X,"R",14X,"Z",5X,45("=")/2X,45("=")/15,3F15.2,2X,3F15.2)
C*****READ PLAST ELLIPSOID LOCATIONS AND SEMIAXES
      P04 FORMAT(LT,1) GO TO 801
      PLAST,2002)(XPL(K),YPL(K),ZPL(K),APL(K),BPL(K),CPL(K),K=1,NPLAST)
      *005 FORMAT("1",59X,"PLAST ELLIPSOIDS")
      WRITE(6,3004)(K,XPL(K),YPL(K),ZPL(K),APL(K),BPL(K),CPL(K),
      1K=1,NPLAST)
      WRITE(6,3005)
      *011 READ DIREC-HIT-NO-KILL ELLIPSOID LOCATIONS AND SEMIAXES
      READ(5,2001)(LT,1) GO TO 802
      READ(5,2002)(XOK,K,1),YOK,K,1),ZOK,K,1),YS(K,1),ZS(K,1),
      1K=1,NDHK)
      *011 READ(46,3006)
      *006 FORMAT(//34X,"DIRECT-HIT-NO-KILL ELLIPSOIDS")
      WRITE(6,3004)(K,XOK,K,2),YOK,K,2),ZOK,K,2),YS(K,2),ZS(K,2),
      1K=1,NDHK)
      *02 READ(5,2001)(XGL(K),YGL(K),ZGL(K),K=1,NGLT)
      WRITE(6,3008)
      *023 FORMAT("1",13X,"GLITTER POINT LOCATIONS")
      *009 READ(5,2002)(K,XGL(K),YGL(K),ZGL(K),K=1,NGLT)
      *009 READ VULNERABLE AREA TABLES
      KMD=6*NCOMP5*NMTS
      DO K=1,NDHK
      LKV=1,MKV
      DO d0=1,KIND
      *03 READ(5,2003)(VUL(F,V,J,K),J=1,11)
      *033 FORMAT(11I7,0)
C*****READ STATIC FREQUENCY INENSITY BY 10000 F INTERVALS

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PROGRAM VNSAR      76/76   OF I=1          FTN 4.6*426      77/09/04. 14.1J.44      PAGE 4

150  STAD(5,2094)STAR
2304 FORMAT(1F10.0)
  0. 8.04  I=1,19
  604  IHE(0,1)=10*(I-1)

165  WRITE(6,3009)THE0,STAR
      3009 FORMAT(1F4.4X,"STATIC FRAGMENT DENSITY BY 10 DEGREE INTERVALS"/
     1."0",1917/1X,134,"1/IX,1967.00
C*****READ FOR EACH MASS CLASS THE FRACTION OF ALL FRAGMENTS BELONGING
C TO THE CLASS AND THE AVERAGE MASS IN GRAINS
KAD(5,2055)(FRAC(I,K),WATE(K),KHT=1,NMIS)
2065 FORMAT(1F10.0)
      WRITE(6,3010)WATE(0),FRAC(0,1),KHT=1,NMIS
3010 FORMAT(//,"FRAGMENT DISTRIBUTION BY HEIGHT"/
     1." AVERAGE WEIGHT      FRACTION OF /"
     2." IN GRAINS      ALL FRAGMENTS"/IX,32," = 1/(F10.1,6,17,2) )
175  C*****READ FUZE CONSTANTS
READ(5,2064)HAFANG,RMAX,DIRE,RORE,DTCE,POPF,POCF
      WRITE(6,3011)HAFANG,RMAX,DIRE,POPF,POCF
      3011 FORMAT(// "HALF-ANGLE OF RADAR FUZE",22X,"=",F6.1," DEGREES"/
     1." MAXIMUM RANGE OF RADAR FUZE",19X,"=",F5.0," FEET"/
     2." DELAY TIME OF RADAR FUZE",22X,"=",F6.3," SECONDS"/
     3." 0.00 PROBABILITY OF RADAR FUZE",17X,"=",F6.3/
     4." DELAY TIME OF CONTACT FUZE",20X,"=",F6.3," SECONDS"/
     5." 0.00 PROBABILITY OF CONTACT FUZE",15X,"=",F6.3)
      HAFANG=HAFANG-POD
C*****READ MISCELLANY
READ(5,2004)RNETH,RMISS,COR,WE
      WRITE(6,3012)VRETH,KMSS,COR,VF
      3012 FORMAT(// "NORMALIZING SPEED FOR VULNERABLE AREA TABLE      =",F7.0,
     1." FPS.      /"
     2." MISS ASSUMED AFTER FAILURE TO FUZE WHEN RANGE =",F5.0," FEET"/
     3." FRAGMENT DRAG COEFFICIENT",21X,"=",F8.5/
     5." FRAGMENT EMISSION SPEED",23X,"=",F7.0," FPS.",/,1")
C*****COMPUTE CONVENTIONAL CONSTANTS
      TH=TA(HAFANG)
      RCHA=RAX*ARCOS(THANG)
C*****INITIALIZE CUMULATIVE COMPONENT SURVIVAL PROBABILITIES
      DO 75 LEV=1,MLEVS
        DO 75 K=1,NCPSS
          CPSS(LEV,K)=1.0
75
C*****INITIALIZE MISSILE
      CALL MISINIT(1)
C*****READ AND PRINT VARIABLE PARAMETERS
      DO 105 I = 1, 5
        NMSPH=I-1
        PFA(5,5,G3) EMIST(JL,JPAFATL),GRPHN(I,J),J=1,3)*(VFAFL,I,J)*
        1. J=1,JL)
105  FORMAT(1A4,216,548,1112,7/16F12.7)
C*****TEST FOR END OF VARIABLE ADDRESSER DATA
      IFENITST = EO. ENDO GO TO 106
      JFARM(I)=JL
115  WRITE(6,6051)I,GRPHN(I,J),J=1,31,JL,JPAFATL,VPAFL,I,J,J=1,JL)
      605 FORMAT(1F4.4,F6.3) EMIST(JL,JPAFATL),GRPHN(I,J),J=1,3)*(VFAFL,I,J)*
      1. J=1,JL)
      SAM  83
      SAM  84
      SAM  85
      SAM  86
      SAM  87
      SAM  88
      SAM  89
      SAM  90
      SAM  91
      SAM  92
      SAM  93

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      1      READ(JA,JAA) , SVLOC(JJ) = 1.5/(C*Y,K*Z,16,7)
      215
      C*****READ AND PRINT EXTR
      106 DO 107 J = 1, 60
      READ(5,505) FPUT(J,K,Z,DNAME
      502 FORMAT(4.14,112,7,6AB)
      C*****TEST FOR END OF REAL EXTRA VARIABLE DATA
      IF(FENDST .EQ. END) GO TO 108
      C*****IF THERE IS EXTRA DATA WRITE PRINTOUT HEADER FIRST TIME ONLY
      IF(J .EQ. 1) WRITE(6,607)
      607 FORMAT(13HOLIST OF EXTR/13H   J   EXTR(JJ))
      EXTR(KK)=Z
      107 WRITE(6,608) K,Z,DNAME
      608 FORMAT(13.1X,1P14.7,3X,6AB)
      C*****READ AND PRINT EXTR
      108 DO 109 J = 1, 30
      READ(5,508) ENOISI,K,IZ,DNAME
      508 FORMAT(4.14,214,6AB)
      C*****TEST FOR END OF INTEGER EXTRA VARIABLE DATA
      IF(FENDST .EQ. END) GO TO 110
      C*****IF THERE IS IEXTRA DATA WRITE PRINTOUT HEADER FIRST TIME ONLY
      IF(J .EQ. 1) WRITE(6,609)
      609 FORMAT(14HOLIST OF IEXTR/13H   J   IEXTR(JJ))
      IEXTR(KK)=IZ
      109 WRITE(6,610) K,IZ,DNAME
      610 FORMAT(3.17,11X,6AB)
      C*****TEST TO SEE IF ACC TRAJECTORY IS GENERATED
      110 IF(IAC .EQ. 2) GO TO 120
      C*****READ AND PRINT STORED STATE VECTOR
      JA=3*1COORD(1)-2
      JA=JA+2
      JA=3*1COORD(2)+6
      JBL=JB+2
      READ(5,510) (SVST(JJ),J=JA,JAA), SVST(JJ),J=JB,JBL
      510 FORMAT(6I12.0)
      C*****CHANGE DEGREES TO RADIANS IF DATA IS IN SPHERICAL COORDINATES
      00 113 I = 1,2
      114 IF(ICORD(1) .EQ. 1) GO TO 113
      K1=6*I-1
      K2=K1+1
      112 SVST(JJ)=SVST(JJ)*RAD
      113 SVST(JJ)=SVST(JJ)*RAD
      C*****CONTINUE
      114 IF(INPARM .EQ. 0) JPARM(1)=0
      00 10 T=1,NPARM
      115 IF(JPARM(T).LT.0) GO TO 30
      10 CONTINUE
      5 CALL COORD(SVST(1),ICORD(1))
      CALL COORD(SVST(1),ICORD(2))
      WRITE(6,612)
      612 FORMAT(3HINITIAL AIRCRAFT STATE VECTOR)
      113 IF(CORD(1) .EQ. 0) WRITE(6,613) SVST
      613 WRITE(6,614) SVST
      245
      256
      266

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PROGRAM CYRCSAM  
 76/74 OCT 1  
 FIN 4.6+4.20  
 17705/4. 14.10.44  
 PAGE 6

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110 READ(5,6/13X,IP=15,6)
  00 20 I=1,NPARM
  1F0 JPARM(I),GE,0) GO TO 20
  00 15 K=1,12
  15 FACTD=SYST(K)
  60 TO 152
*0 CONTINUE
  30 READ(5,511) SSA,ASHMAX,THROTS,ITAU,TAU
  511 FORMAT(3FF.0,11,2F6.0)
  WRITE(6,615) SSA,ASHMAX,THROTS,I,TAU,TAU(I),I=1,ITAU
  615 FORMAT(21HOUT OF SAM SITE,F7.0,4H FT.,2X,15HSTRUCTURE LIMIT,
  1          F4.1,4H G**5.7X,1EMTHROTS SETTING,F5.2?5X,
  2          13NUMBER OF TIME LAGS,I,5Y,17HTIME LAGS IN SEC.,,2F6.3)
C*****TEST FOR READING OF A/C AERODYNAMIC DATA
  1F1ACM *EQ. 01 GO TO 115
  READ(5,512) XAIRC,ACRM,XAIR2
  512 FORMAT(6E12.5,6E12.5)
C*****TEST TO SEE IF A/C DATA IS PROPERLY LOADED AND IS NEEDED
  115 IF(ACM .NE. XAU) GO TO 117
C*****A/C DATA IS NOT LOADED WRITE MESSAGE + READ NEW DATA
  WRITE(6,620)
  620 FORMATT(51H0****A/C AERODYNAMIC TABLES ARE NOT LOADED AND ARE,
  1          14H REQUIRED****15X,17HGOING TO NEXT RUN )
  1          GO TO 100
C*****PRINT OUT TYPE OF A/C TABLES BEING USED
  117 WRITE(6,622) ACM
  622 FORMATT(6H0****,A6,24H IS A/C AERO TABLES****)
  60 TO 125
C*****READ A/C TRAJECTORY DISPLACEMENT AND SAM ALTITUDE AND PRINT THEM
  120 READ(5,515) UP,SSA
  515 FORMAT(6E12.7)
  WRITE(6,625) UP,SSA
  625 FORMAT(4,6H0A/C TRAJECTORY IS READ IN AND IS DISPLACED BY/1P3E15.5/
  1          2NDALITUDE OF SAM SITE ,F7.0)
  121 READ(5,516) UP,SSA
  516 FORMAT(6E12.7)
  WRITE(6,626) UP,SSA
  626 FORMAT(4,6H0A/C TRAJECTORY IS READ IN AND IS DISPLACED BY/1P3E15.5/
  1          2NDALITUDE OF SAM SITE ,F7.0)
  122 READ(5,517) UP,SSA
  517 FORMAT(6E12.7)
  WRITE(6,627) UP,SSA
  627 FORMAT(6H0A/C TRAJECTORY HAS BEEN READ ON UNIT,13.5X,
  1          9THIRE ARE ,14,19H PCINTS IN THE DATA/SX,
  2          16HME DATA TO IS ,24H)
C*****FOR THE TITLE OF NEW A/C DATA
  WRITE(6,628) TACTH,WHT5,FUNID
  628 FORMAT(6H0A/C TRAJECTORY HAS BEEN READ ON UNIT,13.5X,
  1          9THIRE ARE ,14,19H PCINTS IN THE DATA/SX,
  2          16HME DATA TO IS ,24H)
C*****TEST FOR FARADAY VARIATION
  129 IF(NPARM .GT. 0) GO TO 130
C*****WRITE TITLE FOR EDITOR IF NEEDED
  1F1EDIT *LE, 0) GO TO 140
  WRITE(6,629) 7611,TITLE
  7611 FORMAT(6H 1 ,10A6)
  145 JV=0
  60 TO 140
C*****INITIALIZE PARAMETER VARIATION
  124 IF(NPARM .EQ. 4) GO TO 140
  
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	LINE	OPT=1	PAGE
	17703704 • 14.10.44	FIN 4.64420	7
420	J\$=IPARM41 00 129 J=J5,2	SAM	200
	129 IPARM(J)=0	SAM	201
130	J1=0	SAM	202
	J2=1	SAM	203
	J3=1	SAM	204
131	JV=0	SAM	205
	60 TO 136	SAM	206
135	IF(IPARM .EQ. 0) GO TO 36	SAM	207
136	JV=JV+1	SAM	208
	C*****WRITE TITLE FOR RUN IF NEEDED	SAM	209
	IF(UNIT .NE. 0) WRITE(UNIT,702) JV,TITLE	SAM	210
702	FORMAT(1,J2,18A4)	SAM	211
	C*****INCREMENT PARAMETER INDICES	SAM	212
	J1=J1+1	SAM	213
	IF(J1 .LE. IPARM(1)) GO TO 140	SAM	214
140	J1=1	SAM	215
	J2=J2+1	SAM	216
	IF(J2 .LE. IPARM(2)) GO TO 140	SAM	217
	J2=1	SAM	218
	J3=J3+1	SAM	219
	IF(J3 .GT. IPARM(3)) GO TO 360	SAM	220
	140 CONTINUE	SAM	221
	C*****INITIALIZE MAIN OPTIVE	SAM	222
	TE0=OPT0	SAM	223
	TELOT=9.0	SAM	224
	ISTOP=0	SAM	225
	KTRN=0	SAM	226
	C*****INITIALIZE FLAGS	SAM	227
	DO 142 J=1,6	SAM	228
	IMU(J)=FALSE.	SAM	229
142	ALAG54(J)=1	SAM	230
	DO 143 J = 1, 20	SAM	231
143	PFAG(J)=0	SAM	232
	TEOTL=0	SAM	233
	C*****BRANCH TO INITIALIZE A/C	SAM	234
	IF(IAC .EQ. 2) GO TO 147	SAM	235
	C*****INITIALIZE A/C + MAIN DRIVE	SAM	236
	THEOTL=THROTS	SAM	237
	TIME=0.	SAM	238
	IJ=1	SAM	239
	IDX=0	SAM	240
	C*****LOAD STATES	SAM	241
	DO 145 J=1,12	SAM	242
145	RAT(J)=SVE(J)	SAM	243
	DO TO 147	SAM	244
	C*****LOAD PARAMETERS IF NECESSARY	SAM	245
	147 IF (IIPRM .EQ. 0) GO TO 152	SAM	246
	WRITE(6,630)	SAM	247
	E30 FORMATTED LIST OF PARAMETERS	SAM	248
	60 150 J=1,IPARM	SAM	249
	WRITE(6,E32) (PAOEN(K,J),K=1,31),VARRY(J,X(J))	SAM	250
	E32 FORMATTED LIST OF PARAMETERS	SAM	251
150		SAM	252

F706-SAM CYSNSAP      T4/T6      OPT=1      PAGE 4  
 77/08/04. 16.10.44

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    IF(JPARM(J)) .LT. 01 GO TO 146
    F706-JPARM(J)=VPARM(J,JX(J))
    GO TO 150
    149 IF(TIAC .EQ. 2) GO TO 149
    GX=1.
    IF(MOD(-JPARM(J),11,6) .GE. 4) GX=RAD
    RAC(-JPARM(J))=VPARM(J,JX(J))+GX
    GO TO 150
    169 UG(-JPARM(J))=VPARM(J,JX(J))
    170 CONTINUE
    00 40 I=1,12
    40 SW,I,I=FA(L)
    00 45 I=1,NPARM
    IF(JPARM(I),LT.0) GO TO 5
    45 CONTINUE
    152 CONTINUE
    C*****BRANCH AROUND A/C INIT IF TABLES ARE USED
    TIAC .EQ. 2) GO TO 153
    C*****INITIALIZE A/C AERODYNAMICS
    CALL INAFRO
    C*****INITIALIZE A/C STABILITY COORDINATES
    CALL ADIVAD
    C*****INITIALIZE TRAJECTORY STARTING POINT, TIME, AND TIME IF NEEDED
    IFVISOR .LE. 01 GO TO 154
    FUNID(1)=TITLE(1)
    QUNID(2)=TITLE(2)
    HPTS=0
    TSTOP=TIME
    C*****ATANK OUT PRIM WORDS
    153 00 156 KK=1,4
    154 ZWORD(KK)=PLANK
    C*****INITIALIZE TRAJECTORY INTERPOLATION IF NEEDED
    C*****OR INTERPOLATION IF NOT
    IF(TIAC .EQ. 2) CALL INTIK
    CALL OUTPUT
    GO TO 310
    C*****HATH DRIVE ITERATION
    300 CALL SAMDOC
    C*****BRANCH TO CALL POLICY FOR A/C TRAJ. TO BE GENERATED
    C*****OR
    301 IF(TIAC .EQ. 2) GO TO 304
    CALL POLICY
    GO TO 303
    304 CALL INTIAS
    305 CONTINUE
    IF(1 RAIN .GT. 0) CALL MISSILK90H
    TIME=TIME+HM
    C*****STORE A/C TRAJECTORY IF NEEDED
    310 IF(VISOR .LE. 0 .OR. TIME .LT. TSTOP) GO TO 315
    C*****UPDATE POINTER AND TIME TO STORE
    HPTS=HPTS+1
    TSTOP=TSTOP+25
    GO 311 J = 1,
    311 VINTAS(I)=C(I)
  
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PROG, SAM & SASH PAGE 9  
 RTH 4.6+42C 7719.0664. 14.10.44  
 PROG, SAM & SASH PAGE 9  
 RTH 4.6+42C 7719.0664. 14.10.44  
 425 TO 312 J = 1, 4  
 312 REC(J,NPITS)=UNIT(J)  
 C\*\*\*\*\* TURN OFF STORAGE IF NO MORE FROM  
 IF NPITS .EQ. 0, GO TO 150  
 C\*\*\*\*\* WRITE DATA SET FOR PLOT IF DESIRED  
 515 TFPLOT .EQ. 1,0. TIME .LT. TFLOT 60 TO 322  
 TFLOT=TFLOT+DPLOT  
 TFLOT=1#L0UT+1  
 60 320 J=1,6  
 IF MFLAG(J) .LE. 0, GO TO 318  
 IFLAG(J)=TPUF,  
 GO TO 320  
 318 RM3, JI=-1.  
 320 CONTINUE  
 WRITE (10,501) (RA(J),J=1,3), (RM(J,K),J=1,3), K=1,41  
 501 FORMAT(1PBE10.3)  
 322 CONTINUE  
 IF TIME .GE. TOTAL .OR. ISTOP .EQ. 1) GO TO 330  
 IF TIME .LT. TPO) GO TO 309  
 TPO=TIME+DTPO  
 CALL OUTPUT  
 60 TO 300  
 C\*\*\*\*\* END OF ENCOUNTER  
 510 WRITE(6,640) IJ  
 640 FFORMAT(IJ,10HEND OF ENCOUNTER)  
 IJ=0  
 CALL OUTPUT  
 C\*\*\*\*\* WRITE OUT A/C TRAJECTORY IF NEED BE  
 IF TAOUT .GT. 0, WRITE (1AOUT,700) RUMID,NFTS,  
 1 (XT(K,IJ),K=1,9,9G10.7) ,IK=1,9,1L=1,NPITS)  
 C\*\*\*\*\* WRITE CONTROL DATA SET FOR PLOT IF NECESSARY  
 NVEH=1  
 CO 351 J=1,6  
 IF TIME(J) .NEVH+1  
 751 CONTINUE  
 WRITE (9,705) TITLE, TFLOT, NVEH, DFLOT  
 755 FORMAT(18A/I3,11,5H1234,F6.2)  
 GO TO 135  
 C\*\*\*\*\* END OF TASK  
 760 WRITE (6,641)  
 661 FORMAT(15H0TASK COMPLETE)  
 GO TO 100  
 100 STOP  
 END

SUBROUTINE MISSILKAP  
 COMMON/DEFIN/TYPE,HEA,START,TOTAL,EXTR(60),EXTXR(30),ITLE(16),  
 1 CEG,6,P1,RAT  
 COMMON/FLAGS/JPOL,KPOL,LPOL,JSAH,KSAM,LSAM,NMIN,ISTOP,JAC,JEDI,  
 1 IGGUE,IPODE,MFLAG(120),JV  
 COMMON/ATRCF/RA(6),WA(6),ACOM,AGUTA,AQUD,AQUT,VDOU,  
 1 UNITV,VA(3),A(3),A(3),A(3),A(3),D(3),MACH,CA,  
 2 ALPHA,LIFT,DRAG,GEFORCE,UNITL(3),CL,GO,CLMX,SA,CA,  
 2 THRUST,THRUT,ASMAX,ACTTION,DSMAX,DUMPL,DUMCL,IIAU,  
 TAU(2),AD(2,2)  
 4 COMMON/ADATA/THIP(12,21),CD(2(21),MACHMX(13),ALPHAC(21),SAM  
 1 ,DADCL(21),ULMAX(21),AWEA,WEIGHT,SAM  
 COMMON/MISSL/FM(6,20),VM(6,20),WACH(20),OAM(20),SSA,AREAM,  
 ! VM(2,20),DET(2,20),AB(2,20),MN(20),RAN(20),EAM(20) SAM  
 COMMON/PRINJ/LJ,LX  
 FEAL LIFT,MACH,MACHX,MACHMX  
 REAL ANU(2),LX(12,2,20),LX(212,2,20),LY(12,2,20),LY(212,2,20),  
 1 TX(12,2),TY(2(21),FTI(21),RTT(21),AMHS(3,20),CC6(2,20),  
 C01(2,20),CC(42),CCW(21),CD(2(2,20),CD3(2,20),CD(2,20),  
 2 005(2,20),OF(2(2,2,201,601(4),CD(2(6,21,OE(2(2),  
 3 AM(2,20),CHY(26),CHM(2),TL(26),V1(13,20),  
 4 AM(1,3),CME(5,3,201,X184281,X194265,X19120),CX19120),  
 5 P(13,6x2(31,RX3(31),VZ(31),S75(20),C75(20),  
 6 VMS(3,20),SCS(20),V1(22),V2(22),OMES(1),TF(4),  
 7 COMMON /ONE/ ECOMP,SHLEVS,NTS,NSHLS,NDLAST,NGLLT,THA,RCHA,VE,  
 10FL,1,NCM,1CP,ZCH,VHE,THL,1  
 COMMON /TWO/ XGL(25),YGL(25),YGL(25),YSH(25),YSH(25),  
 1 ASH(25),BSH(25),CSH(25),SBL(25),YBL(25),ABL(25),  
 2 CH(125),YCHOL(120),YCHOL(120),ZCHOLD(20),EX(125),WY(125),  
 3 TAM(19),TM1(9)  
 COMMON /THREE/ NODPLY,NGHHR,REHK,HAFNG,PHAY,DIRF,PURF,DCFC,PJCF,  
 4 PMSS,COP  
 COMMON /FOUR/ X0(25),Y0(25),Z0(25),XS(25,2),YS(25,2),  
 175(25,2),WATE(10)  
 COMMON /FIVE/ VOL(2,11,72),STFAP(19),FRACT(10),SLOW(10)  
 COMMON /SIX/ CPS(6,25)  
 EQUIVALENCE (TA(11),ORBIT(11))  
 EQUAL TA(1),PS(4,25),FSF(4,25),FSFF(4,25),CPST(4,25),PKAC(4),  
 1 CFAC(4),KLEV(2,6),READ1(30),HEAD1(30),READ3(3),HEAD4(3),HEAD5(3),  
 2 READ6(3)  
 LOGICAL RAFFUTE  
 EQUIVALENCE(CCX(10),CC1,(CCX(12),CC2),(CCV(11),CC7),(CCV(12),CC8))  
 DATA KLEV,HEAD1,HEAD2,HEAD3,HEAD4,"KILL LEVEL","1","KILL LEVEL",  
 1,"?","KILL LEVEL","?","4","KILL LEVEL","?",  
 ?,"HOT","CUMULATIVE","SURVIVAL","AL",  
 ?,"PROBABILITY","P","ROBABILITY","SURVIVAL","  
 3,"P","ROBABILITY","SURVIVAL","  
 4,"P","ROBABILITY","SURVIVAL","  
 DATA HEAD5,HE ADD,"  
 1,"AIRCRAFT","FT",  
 "KILL","KILL",  
 DATA IF / 4,3,7,4,7,3,6,/,  
 DATA P12 / 6,28316,/  
 DATA HHS / 0,0/  
 SHOTFA(1) \* 2 + SHOT(2) \* 2

FIN 4.5420

PAGE 1

77/06/64. 14.10.44

PAGE 2

77/08/04. 16.10.44

FIN 4.64420

OPT=1

74/74

ROUTINE P1S1T1

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SPR=SCR1(PH0)PA(4)
0D1(1)=(VA(31*QHO-KA(3))+FA(1)*VA(1)+KA(2)*VA(2))/SPR
1*(SPR*FA(4))
0D1(2)=(RA(1)*VA(2))-FA(2)*VA(1)/SPR
0D1(3)=PA(5)
0D1(4)=PA(5)
IF(KFILL .EQ. 0) GO TO 2
00 1 J = 1,2
00 1 I = 1,6
1 UE2(I,J)=001(1)
KFILL=0
60 TO 4
2 00 5 J = 1,2
00 3 I = 1,6
L01(1)=(0D2(1,J)-0D1(1))*I1*DD1(1)
3 0D2(1,J)=0D1(1)
4 IF(IPCOF .EQ. 2) GO TO 5
0D3=(0D3-FA(4)*TT2+FA(4)
Lc3D=0D3-0D3*DD3/HM
0D3S=L03
5 00 1000 I=1,29
IF(MFLAG(1) .LE. 0 ) GO TO 1000
T#=(TIME-TL01)
IF(T# .LT. T#(MFLAG(1))) GO TO 6
MFLAG(1)=MFLAG(1)+1
IF(MFLAG(1) .LT. 5) GO TO 6
WRITE(6,2000) IJ,MNAD,TIME
2000 FORMAT(1I,15H***MISSILE NO.,13,25H SELF DESTROYIS AT TIME =,F0.3)
IF(IFERIT .GT. 0) WRITE(6,5001) JV,(TITLE(J),J=1,2),I,MNAD,
1 TIME,CRH(J,1),J=1,3),(PA(J),J=1,3)
1000 FORMAT(2H 2,I2,2A4,2H 4,2I2,7A4)
IJ=1,J
IJ=0
CALL OUTPUT
MFLAG(1)=0
NMIN=NMIN-1
60 TO 1000
6 IF(IFGUIDE .LT. 6.) GO TO 23
IF(IFRAMID .LE. EXTR(54)) GO TO 822
806 IF(MFLAG(1) .LT. 3) GO TO 7
SP=RAIS1-PH0,T
IF(ARS(SPO) .GE. PI ) SP=SP-SIGN(P12,SP0)
IF(CAP(SPO) .LE. .0671 .AND. ABS(RA(6)-RM(6,1)) .LE.
1.0971) GO TO 7
WRITE(6,2001) IJ,INID,TIME
2001 FORMAT(1I,15H***MISSILE NO.,13,22H REFAKS LOOK AT TIME =,F0.3)
IF(IFERIT .GT. 0) WRITE(6,5001) JV,(TITLE(J),J=1,2),I,MNAD,
1 TIME,CRH(J,1),J=1,3)
5001 FORMAT(2H 2,I2,2A4,2H 4,2I2,7A4)
IJ=1,J
IJ=0
CALL OUTPUT

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SUBROUTINE MISSL      74/74      OPT=1      FIN 4.64420      17/03/04. 14.10.44      PAGE 4  
  
 160  
 CC4=7\*X\*(D1(2))  
 DO 21 J = 1,2  
 X35=2.\*((CC4\*(J))-C01(J,J,11)/HM  
 C06(J,I)=GCC6(J,J)-X335\*TT3\*X17  
 C11(4,J,I)=CC4\*J,D  
 IF(TABS(GCC6(J,J),61,61,  
 21 C0411,NUF  
 C07=C11\*CC6(1,1)\*CC5\*CC5  
 C08=CC2+CC4\*(2,1)+CC4  
 E0 22 J = 1,2  
 C12(CJ,I)=CC2(CJ,1)-CCY(CJ)\*TT4+CCY(CJ)  
 VVS=VM(J,1)  
 VM(J,1)=C02(CJ,1)+C02(CJ,11)-C02(CJ,11)/HM  
 AMU(J,I)=(VM(J,I)-VVS)/HM  
 22 C03(J,I)=C02(J,I)  
 GO TO 23  
 23 RER2=FAM(I)\*\*2  
 ORFS(I)=((FAC(2)-RM(2,J))\*VAC(3,I))-((RA(3)-RM(3,I))\*  
 1 (VA(2)-VM(2,I))/RRR2  
 OME5(I2)=((RA(3)-RM(3,I))\*VAC(1,I))-((RA(1)-RM(1,I))\*  
 1 OME5(I3)=((RA(1)-RM(3,I))\*VAC(1,I))-((RA(1)-RM(1,I))\*  
 1 (VA(1)-VM(1,I))/RRR2  
 1 CP=SOFT(VM(1,I)\*\*2\*VM(2,I)\*\*2)  
 CX=-VM(3,I)\*(VM(4,I)\*CR)  
 VVS=VM(1,I)  
 VM(1,I)=(OPEST1\*VS(2,I)-UMES(21)\*VM(1,I))/CP\*EXTR(53)  
 AMU(I)=VM(1,I)-VVS/HM  
 VM=VM(2,I)  
 VM(2,I)=(C\*COMES(11)\*VM(1,1)\*COMES(21)\*VM(2,1))+VM(1,1)\*COMES(31)  
 1 AMU(2)=(VM(2,I)-VVS)/HM  
 23 CALL ATMO5(I)  
 PVZ=MFLAG(I)  
 GO TO 24,25,25,261, NEXT  
 26 WM=4760.-281.\*TH  
 T2=6.0000.  
 GO TO 27  
 25 WM=2623.-2E+4\*TH  
 T2=6.000.  
 GO TO 27  
 26 WM=1376.  
 T7=0.C  
 27 LBS=MM/G  
 IF(THX2.E0.11.GC.TU.46,  
 1 F1=CMY7.E0.21 GO TO 48  
 711=MACHM(I)  
 TH=1./Z11\*\*2  
 5P2=-578.\*0122\*T8  
 IF(Z11.GT.2.1 GO TO 28  
 UF=-.256\*1.093\*11  
 GM=-.256\*.089\*11  
 FM=-.06565\*.05211\*711

STATE	OUTFILE	INFILE	TAKEOFF	DEPARTURE	PAGE
SIMULATED			747/74	0P1=1	16.10.44
			747/6.6.64.20	77/08/04.	16.10.44
1	CX3=-.02+.135*Z11 60 TO 29				1083
2	CF=-1.536+1.577*Z11 CX1=-.79+1.15*6*Z11 CY2=-.00515+.6329*Z11 CY3=.04+.085*Z11				1089
29	IF(EM .LT. 47.3) GO TO 31 SP2=0.0				1090
	IF(711 .GT. 2.75) GO TO 47 CF=CF-.0475+.15*Z11				1091
	GO TO 31				1092
	CF=CF+.207+.057*Z11				1093
31	CF=CF+.75E-06*(55A8RH11,1D) IF(Z11 .GT. 1.0) GO TO 52 CY1=.601-.468*Z11				1094
	GO TO 35				1095
	IF(Z11 .GT. 2.75) GO TO 33 CY1=.316-.291*Z11				1096
33	GO TO 55				1097
34	IF(Z11 .GT. 2.75) GO TO 34 CY1=-.34				1098
	GO TO 35				1099
35	CY1=-.92+.128*Z11 IF(Z11 .GT. 3.12) GO TO 36 CY2=-.9143-.05865*Z11				1100
	GO TO 37				1101
36	CY2=-.3005-.628*Z11				1102
37	CY3=-5.*CX3 SP3=0AM019*(1116,-3.6*T#1 IF(UMFLAG11 .GT. 3) SP3=0AM011/376. IF(0AM011 .LT. -3.444.1) GO TO 38 IF(0AM011 .GE. .9030.1) GO TO 39 SP1=10.55556-.00316582*JAP01)				1103
	GO TO 40				1104
38	SP1=6.*911 GO TO 46				1105
39	SP1=1.				1106
40	SLOPE=(SP1-.SP15*1D)/WM IF(UM .LT. 7.65) SLOPE=-.625E-03 SP54D=SP1				1107
	FO 45 J0UM1 = 1, 8				1108
	SP6=SP1+SLOPE*SLOPE*LOAT(J0UM1-1) DO 42 J = 1,2				1109
	ANZ=ANY(J,1) AN0U=-YV(J,1)*FLUAT16 DO 41 K = 1,2				1110
	LVI(J,K,1)=YV(0K)*LVI(J,K,1)+TXY1(K)*UYT(J,K,1)-CAN0U- LYT(J,K,1)*FLUAT16(J,K,1)				1111
	LX1(J,K,1)=YV(0K)*LX1(J,K,1)+TXY2(K)*UXT(J,K,1)-CAN0U- LYT(J,K,1)*FLUAT16(J,K,1)				1112
	AB0U=LX1(J,K,1) IF(T2(J,K,1)=T2(J,K,1)*LX2(J,K,1)+TXY2(K)*UXT(J,K,1)-CAN0U- LYT(J,K,1)*FLUAT16(J,K,1)				1113
	AB0U=LX1(J,K,1)				1114

SUBROUTINE PLSSTL OPT1  
 T4/T4 0PT1  
 FIN 4.6\*4.20  
 77/08/05 14.19.44  
 PAGE 6

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4.1 AMT=UX2(J,K,1)  

    DEL(J,I)=SP4*0.00517 *UX1(J,2,1)*0.016 *UX2(J,2,I)*FLOAT(3-2*J) SAM 1144  

    *ONEG(J,1,I)=SP4*0.00516  

4.2 TFLAHS(GELCJ,JD)*GT 10.1 DEL(J,1)=SIGN100.*DEL(J,1,I)  

    CNY(1)=GAB(1,1)*CX1*CX2*ABS(GAB(2,1))  

    CX(2)=(-AE(2,1)*CX1*CX2*ABS(GAB(1,1))-CX3*CEL(2,1))*Z1 SAM 1145  

    CNM(1)=GAB(1,1)*CY1*CY2*ABS(GAB(2,1))+CY3*DEL(1,1)*Z1  

    CDM(2)=(-AB(2,1)*CY1*CY2*ABS(GAB(1,1))-CY3*DEL(2,1))*Z1+SP2*CNY(1) SAM 1146  

    UXV(2)  

    AMV(1,I)=-CAM(I)*CHY(I)/745 SAM 1147  

    ADV(2,I)=GAM(I)*CPW(I)*ZHS SAM 1148  

    AM(I,1)=C17-CAM(I)*C17/ZHS-G*SM(19411)-CMG(2,1)*V1113,I*OME6(3,1)*SAM 1149  

    AM(12)=AM(11)*C17-CAM(I)*C17/ZHS-G*SM(19411)-CMG(2,1)*V1113,I*OME6(3,1)*SAM 1150  

    AM(12)=AM(12,J)=0ME(1,1)+V1114,I+0ME(1,1)*V1114,I  

    AM(13)=G*CX19(I,I)*AKY(1,1)-0ME(6,1,1)*V1112,I  

    0D(13)=SP2*CNM(11)+9d25*0ME(6,2,1)*OME6(1,1) SAM 1151  

    UC(2)=SP3*CMH(2)-9d25*0ME(6,3,1)*OME6(1,1) SAM 1152  

    00 43 J = 1,2  

    0tE(J+1,I)=0tE(J+1,I)+.5*GM*100(J)*CDS(J,1,I) SAM 1153  

4.3 ODS(J,I)=OC(J)  

    SFO=0MEG(3,1)/CX19(I,I) SAM 1154  

    GM(1,I)=X18(I,I)*GM*SP0 SAM 1155  

    X19(I,I)=X19(I,I)*GM*OCE(2,I,I) SAM 1156  

    CX19(I,I)=COS(X19(I,I)) SAM 1157  

    SX19(I,I)=SIN(X19(I,I)) SAM 1158  

    00 46 J = 1,3  

    V111J,I=V11(J,1,I)+.5*GM*(AM1(J,I)+AM1(J,1,I)) SAM 1159  

    AM1(J,I)=AM1(I,J)  

    AM1,I=ATAN(CV11(3,1)/CV11(1,1),ID)*DEG SAM 1160  

4.5 AE(2,1)=ATAN(CV11(2,1)/CV11(1,1))*DEG SAM 1161  

    IF(CA2(1,1)*2*AD(2,1)*2)*LT 90.0,) GO TO 9850 SAM 1162  

    MFLF(6,6005) T,MSNL TIME SAM 1163  

4.605 FORMAT(1.15H***HISTOLE NO.,13,24H GOES UNSTABLE AT TIME =,FB,3) SAM 1164  

    IFFIOT .GT. .90 WAIT(TLE(5000),JV,0) TLE(J,J=1,2),I,MNL(), SAM 1165  

    1 TIME,1*PHU(I),J=1,3),(FAU(I),J=1,3) SAM 1166  

    5004 FORMAT2H 2,12,246,2H 5,212,744) SAM 1167  

    I,J=1,5  

    1,J=0  

    CALL OUTPUT SAM 1168  

    PFLAG,ITI=0 SAM 1169  

    KMIN=KMIN-1 SAM 1170  

    GO TO 4000 SAM 1171  

4.800 CONTINUE SAM 1172  

    SZ=514*(X18(I,I)) SAM 1173  

    U7=CO5*(X16(I,I)) SAM 1174  

    VM1,I,JD=-57*(CX19(I,I)*V11(1,1,1)+SV11(1,1,1)*V11(1,1,1)-SF*V11(2,1,I) SAM 1175  

    VM2,I,JD=5*(CX19(I,I)*V11(1,1,1)+SV11(1,1,1)*V11(3,1,I)-C19(I,I)*V11(3,1,I)) SAM 1176  

    VM3,I,JD=5*(CX19(I,I)*V11(1,1,1)-C19(I,I)*V11(3,1,I)-C2*V11(2,1,I)) SAM 1177  

    GO TO 51 SAM 1178  

4.6 IF(MACHID .NE. 61 .OR. 75 .OR. 97 TO 47 SAM 1179  

    CF=.274/DACB(I,I) SAM 1180  

    GJ TO GJ SAM 1181  

    1182 SAM 1183  

    1184 SAM 1185  

    1186 SAM 1187  

    1188 SAM 1189  

    1190 SAM 1191
  
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STATEMENT	ROUTINE	TYPE	NUMBER	PAGE
			77700704 • 14.10.44	7
324	6.7 CF=4-1.2+1.94*MACH(1)1/MACHE(1)*2 6.0 TO 50	IFN 4.6*4.2C	1192	SAM
4.8 IF (MACHM(1)) .GT. 2.1 GO TO 4.9 CF=4-1.366+1.033*MACHM(1)/MACHE(1)*2 6.0 TO 50	4.9	SAM	1193	
4.9 CF=4-1.336+1.577*MACHM(1)/MACHE(1)*2 AM(1)=CZ-SUM(1)*CF/ZHS-6*5*X19(1) V11(1,I)=V11(1,I)+5*XH4*(AM(1)+AM5(I,I)) AM5(1,I)=AM(1,I) CMEG(2,I)=6.*CMEG(1,I)/V11(1,I),I TF=UTM *UT + 19 - 0.04E6*(2,I)=0. X19(1,I)=X19(1,I)+H*UF6(2,I)	5.0	SAM	1194	
CX19(1,I)=COS(X19(1,I)) SX19(1,I)=SIN(X19(1,I)) VM1,I=CX19(1,I)*CZ/S(1)*V11(1,I) VM2,I=-CX19(1,I)*SZ/S(1)*V11(1,I) VH3,I=SX19(1,I)*V11(1,I)	5.1	SAM	1195	
VH4,I=VH3,I+V11(1,I)	5.2 J = 1,3 RHM,J,RHM(J,I)+H*5*(VH(J,I)*VHS(J,I))	SAM	1196	
VHS(J,I)=VHM(J,I)+H*5*(VH(J,I)*VHS(J,I))	5.2	SAM	1197	
CALC(COMPURM(1,I),I)	5.3	SAM	1200	
XG=R(A(1))-RH(1,I) YCG=R(A(2))-RH(2,I) Z(G)=RA(3)-RM(3,I)	5.4	SAM	1201	
VV=VA(1)-VH(1,I) VV=VA(2)-VH(2,I) V2=VA(3)-VH(3,I)	5.4	SAM	1202	
XG2=-TIA(1)*XG-TIA(2)*YCG-TIA(3)*Z(G) YCG=-TIA(6)*YCG-TIA(5)*YCG-TIA(6)*Z(G) Z(G)=TIA(7)*XCG-TIA(8)*YCG-TIA(9)*Z(G) RAM(1)=S091*YCG*2*YCG*2*Z(G)**2	5.5	SAM	1203	
RAND(1)=-YCG*VX*YCG*YY*Z(G)*V71/FAM(1) IF (MFFLAG(I)) LE 2160 TO 999	5.6	SAM	1204	
IF (HAFFANG .NE. P1/2 .OR. GO TO 5.3 IF (PRA(1),I).GT.0.0 GO TO 5.4 0.*=-RAMD(1)*FAM(1)/(VX**2*YY**2*VZ**2) DET=1.0-DE/X/HM	5.7 G0 TO 56	SAM	1205	
5.3 TMI(1)=CZ*C19(1) TMI(2)=-SZ	5.8	SAM	1206	
TMI(3)=CZ*S19(1) TMI(4)=-SZ*C19(1) TMI(5)=-CZ	5.9	SAM	1207	
TMI(6)=-SZ*S19(1) TMI(7)=SX19(1)	5.9	SAM	1208	
TMI(8)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(3) TIA(6)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9) TIA(5)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1209	
TIA(5)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1210	
TIA(6)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1211	
TIA(5)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1212	
TIA(6)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1213	
TIA(7)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1214	
TIA(8)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1215	
TIA(9)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1216	
TIA(5)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1217	
TIA(6)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1218	
TIA(7)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1219	
TIA(8)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1220	
TIA(9)=TMI(2)*TIA(1)+TIA(5)*TIA(7)*TIA(9)	5.9	SAM	1221	

SUBROUTINE PSS1L 74/74 OPT=1  
 F77 4.6+420 77708/04. 14.10.46 PAGE 8  
 100 SLOWKWT=CUR\*2.0/QA/(AREA\*VAL1\*\*2\*WATE(KWT))\*\*1.0/3.011  
 101 IF(0.0.LT.KWT) GO TO 65  
 102 CALL STAB(X0,Y0,Z0,X5,Y5,Z5,NHMK,T1)  
 103 IF(T1.LT.1.0) GO TO 57  
 104 IF(IRA=0.0).GT.0.0 GO TO 999  
 105 IF(PAP=1.0).LT.RMISS1 GO TO 999  
 106 DO 56 K=1,NCOMPS  
 107 DO 56 LEV=1,NLEVS  
 108 PSILEV,K=1,0  
 109 RAUFLUE=.FALSE.  
 110 GO TO 63  
 111 TBCF=T1+DTBCF/HM  
 112 CALL STAB(X0(1,2),Y0(1,2),Z0(1,2),X5(1,2),Y5(1,2),Z5(1,2),NHMK,12)  
 113 IF(T12.LT.1.0) GO TO 66  
 114 IF(PDCCF.GE.1.0) GO TO 55  
 115 CALL SPAYTBKF,PSCF1  
 116 DO 59 K=1,NCOMPS  
 117 DO 59 LEV=1,NLEVS  
 118 PSILEV,K=PDCCF\*(1.0-PDCCF)\*PSCF(LEV,K)  
 119 RAUFLUE=.FALSE.  
 120 GO TO 63  
 121 IF(112.GT.TBCF) GO TO 63  
 122 DO 62 K=1,NCOMPS  
 123 DO 62 LEV=1,NLEVS  
 124 PSILEV,K=0.0  
 125 RAUFLUE=.FALSE.  
 126 GO TO 63  
 127 IF(PDCCF.GE.1.0) GO TO 61  
 128 CALL SPAYTBKF,PSCF1  
 129 DO 64 K=1,NCOMPS  
 130 DO 64 LEV=1,NLEVS  
 131 PSILEV,K=(1.0-PDCCF)\*PSCF(LEV,K)  
 132 RAUFLUE=.FALSE.  
 133 GO TO 63  
 134 TBCF=DEL1+DTBCF/HM  
 135 CALL STAB(X0,Y0,Z0,X5,Y5,Z5,NHMK,T1)  
 136 IF(T11.LE.1.0) GO TO 68  
 137 CALL SPAYTBKF,PSCF1  
 138 DO 67 K=1,NCOMPS  
 139 DO 67 LEV=1,NLEVS  
 140 PS4LEV,K=PDRCF\*(1.0-F4RF)\*PSCF(LEV,K)  
 141 RAUFLUE=.TRUE.  
 142 GO TO 63  
 143 TBCF=11.0\*TCF/HM  
 144 CALL STAB(X0(1,2),Y0(1,2),Z0(1,2),X5(1,2),Y5(1,2),Z5(1,2),NHMK,12)  
 145 IF(T12.LT.1.0) GO TO 74  
 146 IF(DC1.GE.1.0) GO TO 74  
 147 SAMI 106  
 148 SAMI 109  
 149 SAMI 190  
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 154 SAMI 195  
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 197 SAMI 238  
 198 SAMI 239  
 199 SAMI 240



SUBROUTINE MISSILE      P4/P4      OPT=1

FTN 4.6+42G      77/68/04. 14.10.44

PAGE 10

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      DO 84 J=1,2
      F(X1(J))=R(A(J))-0.6X*VAL(J)
      RX2(J)=RM(J,1)-0.6X*VM(J,1)
      R(X3(J))=RX2(J)-F(X1(J))
      IF(IAC.EQ.2) CALL INBL
      CALL COMMUNIT(Y,X)
      CALL COU(LA1,RX3,Y)
      CALL LOT(LA1,RX3,Z)
      X:=SCRT((RX3(1)+2*RX3(2))*2+RX3(3))*2
      A01=ATAN2(Y,X)*0.6E6
      X02=ASIN(Z/X01)*DEG
      XX=-TP1(1)*RX3(1)-TM1(4)*RX3(2)-TM1(7)*RX3(3)
      YY=-TP1(2)*RX3(1)-TM1(5)*RX3(2)-TM1(8)*RX3(3)
      ZZ=-TP1(3)*RX3(1)-TM1(6)*RX3(2)-TM1(9)*RX3(3)
      X01=ATAN2(-Y,X)*0.6E6
      X02=ASIN((-Z/X01)*DEG
      WRITE(6,4000)A01
      IF(RAUFUZE) WRITE(6,2021)M(1),PSH,X00,RX1,(VAL(J,J=1,3)),RX2,
      1(VH(J,J=1,3)),Y,Z,X02,XB1,XX,VY,ZZ,XXB2,XXB1
      2702 FORMAT(1," *** MISSILE NO.",I3," BURSTS DUE TO RADAR FUZE AT TIME=",,
      1FA,3," AND RANGE =",F10.1," FLEET",
      2" STATE VECTORS IN INCITAL COORDINATES AT PURSUIT",I3X,"X",
      3X,"Y",9X,"Z",9X,"VX",9X,"VY",9X,"VZ",/"/ AIRCRAFT",F8.0,
      4,2F10.0,3F10.1," MISSILE",F9.0,2E10.0,3F10.1/
      5"MISSILE IN A/C BODY COORDINATES",6X,"X",8X,"Y",8X,"Z",
      6"X","ELEVATION",AZIMUTH"/3F9.2,F10.2,F12.2/
      7"0A/C IN MISSILE BODY COORDINATES",6X,"X",8X,"Y",8X,"Z",
      8"Y","ELEVATION",AZIMUTH"/3F9.2,F10.2,F12.2,
      IF(KALFUZE) WRITE(6,2031)OME(G,J,1),AH1(J,1),ANY(J,1),DEL(J,1),
      1AF(J,1),J=1,21,OME(G,13),AM1(13)
      2603 FORMAT(1,"MISSILE STATUS AT 00RS",I4X,"0MEG",9X,"AH1",9X,"ANY",
      19X,"DFL",9X,"AB",Z/1P5E12,J1,1P2E12,J1)
      IJ=0
      IF(I,NCT,RAFFUZE) CALL OUTPUT
      WRITE(6,4001)TILEV(J,LEV),J=1,2),LEV=1,NLEV$)
      4001 FORMAT(1," ",4(19X,A1,A2))
      4002 WRITE(6,4002)TILEV(J,LEV),J=1,3),LEV=1,NLEV$)
      4003 WRITE(6,4003)TILEV(J,LEV),J=1,3),LEV=1,NLEV$)
      4004 FORMAT(1," CC4PONENT",4(1X,3A10))
      4005 WRITE(6,4005)TILEV(J,LEV),J=1,3),LEV=1,NLEV$)
      4006 FORMAT(1," ",4(1X,3A10))
      4007 WRITE(6,4007)TILEV(J,LEV),J=1,3),LEV=1,NLEV$)
      4008 FORMAT(1," K",1,NCOMP$)
      4009 LEV=1,NLEV$)
      4010 CFSLEV,K1=CFSLEV,K1*PS(LEV,K1)
      4011 CFSLEV,K1=CFSLEV,K1
      4012 WRITE(6,4006)K,PS(LEV,K1),CFSLEV,K1,LEV=1,NLEV$)
      4013 WRITE(6,4007)K,PS(LEV,K1),CFSLEV,K1,LEV=1,NLEV$)
      4014 TFCND(3LY,LE,C1 GO TO 89
      4015 L:NGE?*NGEOL Y-1

```

SUBROUTINE MISSLT  
 FTN 4.6+4.20  
 77/08/04. 14.10.44  
 PAGE 11

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      DO 87 LEV=1,NLEVS
      DO 87 L=1,LEN0
      PSLEV,L)=PS(LEV,L)+PS(LEV,L+1)-PS(LEV,L)*PS(LEV,L+1)
      PSLEV,L+1)=1.0
      CPSI(LEV,L)=CPSI(LEV,L)+CPSI(LEV,L+1)-CPSI(LEV,L)*CPSI(LEV,L+1)
      87 CPSI(LEV,L+1)=1.0
      88 DO 89 LEV=1,NLEVS
      PSAC=1.0
      CPSAC=1.0
      89 89 K=1,NCNMP5
      PSAC=PSAC*PS(LEV,K)
      CPSAC=CPSAC*CPSI(LEV,K)
      FKAC(LEV)=1.0-PSAC
      C1-KA(LEV)=1.0-CP5AC
      MFITE(6,4,005) ((HEAD4(J),J=1,3),LEV=1,NLEVS)
      WRITE(6,4,001) ((KLEV(J),LEV),J=1,2)+LEV=1,NLEVS
      MFITE(6,4,002) ((HEAD5(J),J=1,3),LEV=1,NLEVS)
      MFITE(6,4,002) ((HEAD4(J),J=1,3),LEV=1,NLEVS)
      MFITE(6,4,002) ((HEAD6(J),J=1,3),LEV=1,NLEVS)
      MFITE(6,4,002) ((HEAD5(J),J=1,3),LEV=1,NLEVS)
      MFITE(6,4,002) ((HEAD4(J),J=1,3),LEV=1,NLEVS)
      MFITE(6,4,002) ((HEAD6(J),J=1,3),LEV=1,NLEVS)
      MFITE(6,4,007) (PKAC(LEV),LEV=1,NLEVS)
      MFITE(6,4,007) (PKAC(LEV),LEV=1,NLEVS)
      6,007 FORMAT(8.4F14.3,F17.3)
      MFITE(6,4,002) ((HEAD4(J),J=1,3),LEV=1,NLEVS)
      LJ=1
      LJY=0
      MFLAG(I)=0.0
      NMIN=NMIN-1
      IF(I>0,I,0) WRITE(6,011),50021,JV,(TITLE(JJ),JJ=1,2),I,MFLG(I),
      11SM,FV2,RY1,XN0,XN1,XN2,FKAC
      5002 F09 A10* 2",12,2A4," 2",*212,14A4,
      999 XCHOL(C(I))=YCP
      YCHOLE(C(I))=YCM
      ZC40L(C(I))=ZCM
      1020 CONTINUE
      RETURN
      END! MISINT
      WRITE(6,3000)
      1000 FORMAT(125H*****MISSILE TYPE *****)
      1700 IF(IH .EQ. HM$) RETURN
      NM5=0
      AFHM=2.01
      TT1=LAP(-HM*10.)
      TT2=EXP(-HM*2.)
      TT3=EXP(-HM/2.)
      TT4=EXP(-HM)
      GM=HM*.125
      TX1(1)=EXP(-GM*10.)
      TX1(2)=TX1(1)
      TX2(1)=EXP(-GP/.10)
      TX2(2)=TX1(1)
      TX1(1)=1.-TX1(1)
      TX1(2)=TX1(1)
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SUBROUTINE MISS1    T4/T4    OPT=1    FIN 4.64420    17/08/04 • 14.10.44    PAGE 12  
 e.05      R111(1)=R111(1)  
 R111(2)=R111(2)  
 TX2(1)=1.0-TX2(1)  
 TX2(2)=TX2(1)  
 R112(1)=+10/GF  
 R112(2)=R111(2)  
 HHS=HF  
 RETURN  
 ENTRY LAUNCH  
 DO 1500 J = 1, 20  
 IF (HFLAG(J) .LE. 0 ) GO TO 1601  
 1500 CONTINUE  
 K6=0  
 RETURN  
 E.01   K=J  
 DO 1502 L=1,6  
 1502 RM(L,K)=0.  
 VM(4,K)=10.  
 V11(1,K)=10.  
 V11(2,K)=0.  
 V11(3,K)=0.  
 PACHM(K)=0.0  
 RH0=RAT(1)\*2.\*RA(2)\*2  
 VH15(K)=FA15)+7.0\*(RA(1)\*VA(1)-RA(2)\*VA(2))/FH0  
 VH(6,K)=+2E2\*5.92\*(RA(6)+7.\*((VA(3)\*PHO-RA(3)\*(RA(1)\*VA(1)+VA(2)))\*VA  
 1(2))+((PHO\*PA(4))\*2))  
 IF (A5\*VM(5,K) .LT. PI) VM(5,K)=VM(5,K)-SIGN(2.\*PI, VM(5,K))  
 CALL COORD(VM(1,K),2)  
 MFLAG(K)=1  
 RAM(K)=PA(4)  
 RAM0(K)=0.  
 NMIN=NMIN+1  
 NX5=NX5+1  
 MU(K)=NX5  
 IF (L<LT .GT. 0) WRITE(1,E017,9993) JV, (TTLF(J),J=1,2), J, MU(L),  
 1  
 E013 FORMAT(2H 2,T2,2A4,2H 1,2I2,4A4)  
 e.06      X19(K)=VM(6,K)  
 X19(K)=SIN(X19(K))  
 CX19(K)=COS(X19(K))  
 CZ5(K)=SIN(X18(K))  
 CZ5(K)=COS(X18(K))  
 90 1503    JI=-.2  
 DEL(J1,K)=0.0  
 W(J1,K)=0.0  
 AIW(J1,K)=0.0  
 CP1(J1,K)=0.  
 CP2(J1,K)=0.  
 CD3(J1,K)=0.  
 AF(J1,K)=0.0  
 OFS(J1,K)=0.0  
 CC(J1,K)=0.  
 SAM 1283  
 SAM 1284  
 SAM 1285  
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 SAM 1333  
 SAM 1334  
 SAM 1335

Statement No.	Line	FTN 4.6+4.70	7710804. 14.10.44	PAGE	1.3
640	1	CU2(J1,K)=0.		SAM	1336
	2	DO 1504 J2=1,2		SAM	1357
	3	BF2(J1,J2,K)=VM(7-J1,K)		SAM	1338
	4	LY1(J1,J2,K)=0,0		SAM	1339
	5	LY2(J1,J2,K)=0,0		SAM	1340
	6	LX1(J1,J2,K)=0,		SAM	1341
	7	LX2(J1,J2,K)=0,		SAM	1342
	8	T1(K)=TIME		SAM	1343
	9	LO 1506 J = 1,3		SAM	1344
	10	VNS(J,K)=VM(J,K)		SAM	1345
	11	AM1SA(J,K)=0.		SAM	1346
	12	AM2G(J,K)=0.		SAM	1347
	13	KA=K		SAM	1348
	14	IF (NMIN .GT. 1) RETURN		SAM	1349
	15	LO3=PA(6)		SAM	1350
	16	DO 155=LO3		SAM	1351
	17	KFTLL=1		SAM	1352
	18	KE TURK		SAM	1353
	19	END		SAM	1354
645	1				
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SUBROUTINE SPHAY(LLT,PSC)  
 COMMON /DRIVE/ TIME,MM,ISTART,TOTAL,EXTR(60),TEXTR(30),TITLE(100),  
 1 DFG,G,PI,PA0  
 COMMON /MISSL/ RM(6,20),VN(6,20),MACHM(420),QAM(20),SSA,AREAM,  
 2 VV(12,20),DEL(2,20),AB(2,20),MN(20),RAR(20),RAP(20),  
 COMMON /FLAGS/ JFOL,KFUL,JSAM,KSAM,LSAM,NMIN,ISTOP,IAC,IEDIT,  
 3 IGO10F,IMODE,MFLAG(120),JV  
 COMMON /ATRC/ RA(61),VA(16),DUM(16),TIA(9)  
 COMMON /ONE/ NCOMPS,NLEVS,NHVS,NSHLD5,NBLAST,MGLIT,THA,RCHA,VE,  
 4 DELT,XCM,YCM,ZCM,VNETHI,I  
 COMMON /TMO/ XGL(25),YGL(25),XSH(25),YSH(25),ZSH(25),  
 5 IASH(25),BSH(25),CSH(25),XUL(25),YUL(25),ZBL(25),ABL(25),  
 6 ZCL(25),XCMOL(120),YCMOL(120),ZCMOLD(120),FX(25),MY(25),ZLE(25),  
 7 TIA(9),TM1(9)  
 COMMON /FIVE/ VUL(2,11,72),SIEAR(19),FRACT(10),SLOW(10)  
 DIMENSION PSC(4,25),SUM(4),VI(3),VS(3)  
 8 DELTA=0.5\*VA(4)  
 XSTAR=XCMOLD(1)+DLT\*(XCM-XCMOLD(1))  
 YSTAR=YCMOLD(1)+DLT\*(YC-M-YCMOLD(1))  
 ZSTAR=ZCMOLD(1)+DLT\*(ZCM-ZCMOLD(1))  
 IF(UNBLAST,LT.1) GO TO 102  
 CO 101 M=1,NBLAST  
 10 IF((XSTAR-ZBL(1))/AEL(M))+2\*((YSTAR-YBL(M))/BBL(M))+2+  
 1 ((ZSTAR-ZBL(1))/CBL(M))+2.GT.1.0, GO TO 101  
 100 DO 100 K=1,NCOMPS  
 100 LEV=1,NLEVS  
 PSC(LEV,K)=0.0  
 RETURN  
 101 CONTINUE  
 102 VT(1)=TIA(1)\*VA(1)+TIA(2)\*VA(2)+TIA(3)\*VA(3)  
 VT(2)=TIA(4)\*VA(1)+TIA(5)\*VA(2)+TIA(6)\*VA(3)  
 VT(3)=TIA(7)\*VA(1)+TIA(8)\*VA(2)+TIA(9)\*VA(3)  
 VS(1)=TIA(1)\*VM(1,1)+TIA(2)\*VM(2,1)+TIA(3)\*VM(3,1)  
 VS(2)=TIA(4)\*VM(1,1)+TIA(5)\*VM(2,1)+TIA(6)\*VM(3,1)  
 VS(3)=TIA(7)\*VM(1,1)+TIA(8)\*VM(2,1)+TIA(9)\*VM(3,1)  
 VT(1)=VT(1)\*VS(1)+VT(2)\*VS(2)+VT(3)\*VS(3)  
 A=VA(4)\*2\*VM(4,1)\*2\*VE\*2.0\*VIVS  
 VKAT=VE\*VM(4,1)  
 VKAT2=VRAT\*2  
 NO 31 K=1,NCOMPS  
 CX=FX(K)-YSTAF  
 30 CY=HY(K)-YSTAC  
 CZ=ZEFF(K)-ZSTAR  
 B=2.0\*(DX\*(VT(1))-VS(1))+UY\*(VT(2))-VS(2)+DZ\*(VT(3))-VS(3))  
 C=DX\*\*2+DY\*\*2+DZ\*\*2  
 D=0\*B-4,0\*A,C  
 IF(0.6E+0.0) GO TO 3  
 1 DO 2 LEV=1,NLEVS  
 2 PSC(LEV,K)=1.0  
 40 GO TO 31  
 3 T1=(E-SORT(0))/((2.0\*A)  
 T2=(-E-SORT(0))/((2.0\*A)  
 IF(T1.GE.0.0) GO TO 6  
 50

SUBROUTINE	SFTAY	74/74	OPT=1	FTN 4.6+4.20	77/08/04 • 14.10.44	PAGE
55	IF(T12.LT.0.0) GO TO 1 TEST=12 GO TO 6	SAH	1989			
60	4 IF(T2.GT.0.0) GO TO 5 TEST=11 GO TO 6	SAH	1990			
65	5 TEST=AMIN1(G1,T2) 6 DO 7 LEV=1,NLEV\$ 7 SUMLEV=0.0 DO 28 KWT=1,NKWT KOUNT=0	SAH	1991			
70	8 FLX=D*X*VT(1)*TEST ELV=DY*VT(2)*TEST FLZ=0.2*VT(3)*TEST FL=SURFELX**2+ELY**2+ELZ**2 KOUNT=KOUNT+1 DELTAX=ELX/EL DELTAY=ELY/EL DELTAZ=ELZ/EL	SAH	1992			
75	ELDOUT=BETAX*VT(1)+BETAY*VT(2)+BETAZ*VT(3) CUSGAM=(BETAX*VS(1)+BETAY*VS(2)+BETAZ*VS(3))/VM(4,I) VZERO=M(4,I)*COSGAM+SOR((VM(4,I)*COSGAM)**2+VE**2-VH(4,I)**2)	SAH	1993			
80	F=SLOW(KWT)*EL-ALOG(1.0+SLOW(KWT)*VZERO*TEST), VZERO=VZERO*(VM(4,I)-ELDOUT*VM(4,I)*COSGAM)/ 1.0L*(VZERO-VM(4,I)*COSGAM) FPRI=F*SLOW(KWT)*(ELDOUT-(VZERO*COSGAM)/ 1.0L+SLOW(KWT)*VZERO*TEST)) IF(FPRIME.GE.0.0) GO TO 28 TNFM=TEST-F/FPRI IF(THMW.LT.0.0) GO TO 28 IF(CABS(THMW-TEST).LE.DELTA) GO TO 10	SAH	1994			
85	IF(KOUNT.GT.10) GO TO 28 9 IF(ST=INEW GO TO 8	SAH	1995			
90	10 SPOWN=EXP(-SLOW(KWT)*FL) COSTHT=VZERO*COSGAM-VH(4,I)/VF IF(CABS(COSTHT).GT.1.0) COSTHT=SIGN(1.0,COSTHT)	SAH	1996			
95	11 VIG=VT(1)*TEST VIG=VT(2)*TEST ZTG=VT(3)*TEST IF(NSHLDS.LT.1) GO TO 15 IF(BETAX.EQ.0.0) GO TO 9 DO 14 L=1,NSHLDS IF(TEX(K).NE.XSH(L)) GO TO 11 IF(YH(K).NE.YSH(L)) GO TO 11 IF(ZEEL(K).EQ.7SH(L)) GO TO 14	SAH	1997			
100	11 0X=XSH(L)+YTG-XSTAR/A SH(L) 0Y=YSH(L)+YTG-YSTAR/B SH(L) 0Z=(ZSH(L)+ZTG-ZSTAR)/CSH(L) TAY=(BETAY*ASH(L))/(BETAX*B SH(L)) TXZ=(FPETAZ*ASH(L))/(FPETAX*C SH(L)) A=1.0*XY**2+TXZ**2 F=-2.0*(DX+1*XY+1*Y+1*Z+1)	SAH	1998			
105		SAH	1999			
		SAH	2000			
		SAH	2001			
		SAH	2002			
		SAH	2003			
		SAH	2004			
		SAH	2005			
		SAH	2006			
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		SAH	2026			
		SAH	2027			
		SAH	2028			
		SAH	392			
		SAH	2030			
		SAH	2031			
		SAH	2032			
		SAH	2033			
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		SAH	2035			
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		SAH	2038			
		SAH	2039			
		SAH	2040			
		SAH	2041			

74/76 OPT=1  
 77/08/04. 14.10.44  
 FTN 4.6•4.20  
 PAGE 3  
 SMCOUTINE SPAY  
 C=DX\*\*2+DY\*\*2+DZ\*\*2-1.0  
 D=0\*B-4.0\*A\*C  
 IF Q.L.E.0.01 GO TO 14  
 C=SORT(0)  
 00 13 LL=1,2  
 X0LM=(-3+D\*(-1.0\*\*LL))/((2.0\*A))  
 XPUNCH=XSTAR\*X0LM\*ASH(L)  
 IFXPUNCH.LE.XSTAR GO TO 12  
 IFEX(KI)\*XIG-XSTAR.GT.XPUNCH) GO TO 20  
 GO TO 13  
 12 IF EX(KI)\*XIG-XSTAR.LT.XPUNCH) GO TO 26  
 13 CONTINUE  
 14 CONTINUE  
 15 IHELT=ATAN2(SORT(1.0-COSTHT)\*\*2,COSTHT)  
 KANG=18.0\*THE(TAP)+1.0  
 IF(KANG-N.E.19) GO TO 16  
 STEAR=STEARI9  
 GO TO 17  
 16 CTR=STEAR(KANG)+STEAR(KANG+1)-STEAR(KANG+1)\*  
 1/18.0\*THETA/PT+1.0-KANG)  
 17 VWHITE=VZERO\*SDOWN  
 VNET=SQRT((VWHITE\*ULIX-VT(11))\*\*2+(VWHITE\*BTAY-VT(2))\*\*2+  
 1/WHITE\*\*2\*VTAZ-VT(39)\*\*2)  
 KV=10.9\*VNED/VNETHT1.0  
 SFREAD=VRAT?\*ABS(VRAT+COSTHT)/(VRAT2\*2.0\*VRAT\*COSTHT+1.0)\*\*2+  
 DO 27 1XYZ=1,\*  
 IF(IXYZ-2118,20,22  
 18 SHAU=A3S\*PBTAY1  
 IF(BETAX.GE.0.0) GO TO 19  
 19 IFACE=0  
 GO TO 24  
 20 SHAD=ANS(BETAY)  
 IF(BETAY.GE.0.0) GO TO 21  
 IFACE=3  
 GO TO 24  
 21 IFACE=2  
 GO TO 24  
 22 SHAD=ANS(BETAZ)  
 IF(BETAZ.GE.0.0) GO TO 23  
 IFACE=4  
 GO TO 24  
 23 IFACE=5  
 24 KFACE=6\*K-IFACE  
 JINDEX=X=NWTS\*(KFACE-1)\*KWT  
 DO 26 LEV=1,NLEV  
 IF KV.LT.111 GO TO 25  
 VAREV=VUL(LEV,1,JINDEX)  
 GO TO 26  
 25 VAREV=VUL(LEV,KV,JINDEX)\*(10.0\*VNETHI+1.0-KV)\*  
 1\*(VUL(LEV,KV+1,JINDEX)-VUL(LEV,KV,JINDEX))  
 26 SHM(L,V)=SUP(LEV)\*STIR\*FRAC1(KWT)\*VAREV\*(SPRFAD\*EL\*\*2)

4  
PAGE  
77/08/04. 14.10.44  
FTN 4.6+420  
SUBROUTINE STAY T4/T4 OPT=1  
160  
27 CONTINUE  
28 CONTINUE  
DO 30 LFV=1,NLEVS  
IF (SUM(LFV).LT.9.0) GO TO 29  
PSC(LFV,KI)=0.0  
GO TO 30  
29 PSC(LFV,KI)=EXP(-SUM(LFV))  
30 CONTINUE  
31 CONTINUE  
RETURN  
END  
165  
170

SUBROUTINE STAR

7474

OPT=1

77/08/04 14.10.44

PAGE 1

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1      SUBROUTINE STAR(X0,Y0,Z0,X5,Y5,Z5,N,DLT)
      COMMON YMISSL,KM6,2U,VH16,201,MACH(120),QAM(120),SSA,AREAM,
      1V(12,201),DELT12,201,AE(12,201,MN(201),RA(120),FAD(201),
      COMMON /FLAGS/ JPOL,KSAM,LSAM,NMIN,ISTOP,IAC,IEUT,
      1
      1 IF(GD0,E,IMODE,IMFLAG(120),JV
      COMMON /ATIFCF/ DUH(128),TIA(19)
      COMMON /ONE/ NCOMPS,NLEVS,NMITS,NSHLCS,NPLAST,NGLIT,THA,RCHA,WE,
      1OELT,XCH,YCH,ZCH,VNEHTI,I
      COMMON /TWO/ XGL(125),YGL(125),ZGL(125),XSH(125),ZSH(125),
      1ASH(125),BSH(125),CSH(125),XBL(125),YBL(125),ZBL(125),AUL(125),BBL(125),
      2CBL(125),XCMLD(120),YCMLD(120),ZCMLD(120),EX(125),HY(125),ZEE(125),
      3TAM(9),THI(9)
      DIMENSION X0(125),Y0(125),Z0(125),XS(125),YS(125),ZS(125),
      DLT=2.0
      IF(10,LT,1) RETURN
      OX=XCMOLD(1)-XCH
      OY=YCMOLD(1)-YCH
      OZ=ZCMOLD(1)-ZCH
      DO 4 J=1,N
      4 F14DZ,E0.0,0.0 GO TO 1
      XY2=(XS(1,J)*VS(J))**2
      XZ2=4*XS(1,J)*ZS(J)**2
      YZ2=(YS(1,J)*ZS(J))**2
      YZ=XCMLD(1)-YCMLD(1)+YCMLD(1)*7CH+XO(J)*DZ
      YZ=YZ+YCMLD(1)*YCMLD(1)*ZCH+YO(J)*DZ
      A=XY2*DZ**2*XZ2*0Y**2*Z**2*DZ*DZ**2
      B=-2.0*(DX*YZ*1XZ*DY*XZ2*AYZ*DZ**2*XY2*ZO(J))
      C=XY2*YZ**2+YZ2*T*Z**2*XY2*Z*2*YZ*2*ZO(J)*2*TS(J)**2
      E=A-B-4.0*A*C
      IF(10,LT,0,0) GO TO 4
      0=SORT(0)
      Z5=(-E*SIGN(D,D))/((2.0*A)
      DFLL=(ZCMOLD(1)-Z3)/DZ
      GO TO 3
      1 IF(DY,E0,0,0) GO TO 2
      A=(YS(1,J)*DX)**2*(XS(J)*DY)**2
      TXY=XCH*YCMLD(1)-YCMLD(1)*YCH+XO(J)*DY
      XY2=(XS(1,J)*DY)**2
      B=2.0*(YS(1,J)*2*0X*IXY-XY2*YO(J))
      C=(YS(1,J)*TXY)**2*XY2*(YO(J)**2+YS(1,J)**2)*
      1((IZCH-Z0(J))/ZS(J))**2-1.0,
      D=9*B-4.0*A*C
      IF(10,LT,0,0) GO TO 4
      0=SORT(0)
      Y3=4.-D*SIGN(D,D)/(2.0*A)
      ULL=(YCMLD(1)-Y3)/DY
      GO TO 3
      2 IF(DX,E0,0,0) RETURN
      1YZ=1.2-(IZCH-Z0(J))/ZS(J))**2-(YCH-YO(J))/VS(J))**2
      IF(IYZ,LT,0,0,1,0 TO 4
      YZ=SORT(IYZ)
      X3=AO(J)*XS(J)*SIGN(IYZ,DX)
      RLL=(XCMLD(1)-Y3)/DY

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51

SUBROUTINE	STAB	TR/TA	OPT=1	FTN 4.6+4.20	77/08/64.	14.10.44	PAGE	2
3	IF(DELL,LT,0.0)	GO TO 4		SAM	2159			
	IF(DELL,LT,0.0)	0LT=0FL		SAM	2160			
4	CONTINUE			SAM	2161			
	RETURN			SAM	2162			
	END			SAM	2163			

SUBROUTINE CCNUZTE  
 FTN 4.64420  
 77/06/04. 14.10.44  
 PAGE 1

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1      SUBROUTINE CCNUZTE
      COMMON /MISSL/ RM(6,20), VM(6,20), MACMM(20), OMM(20), SSA, APFAH,
      1 W(2,20), DEL(2,20), AB(2,20), MN(20), RAM(20), RAPD(20)
      COMMON /FLAGS/ JPUL, KFJL, JSAM, KSAM, LSAM, NMN, ISTOP, IAC, IEDIT,
      1 IGUIDE, THODE, MFLAG(20), JV
      COMMON /AIRCFC/ DUM(20), TIA(9)
      COMMON /ONE/ PICOMPS, NLEVS, NMHS, NSHLDS, NBLAST, NGLIT, THA, RCHA, VE,
      1 CCL, XCH, YCH, ZCH, VNE, THI, I
      COMMON /TWO/ XGL(25), YGL(25), ZGL(25), XBL(25), YBL(25), ZBL(25),
      1 ASH(25), BSH(25), CSH(25), XBL(25), YBL(25), ZBL(25), ABL(25), BBL(25),
      2 CHL(25), XCMOLD(20), YCMOLD(20), ZCMOLD(20), EX(25), WY(25), ZEE(25),
      3 TAM(9), THI(9)
      REAL XG(2), YG(2), ZG(2)
      DELT=2.0
      DO 10 IGL=1,NGLIT
      T1=XGL(1GL)-XCMOLU(1)
      T2=YGL(1GL)-YCMOLU(1)
      T3=ZGL(1GL)-ZCMOLU(1)
      XG(1)=TAM(1)*T1*TAM(2)*T2*TAM(3)*T3
      YG(1)=TAM(4)*T1*TAM(5)*T2*TAM(6)*T3
      ZG(1)=TAM(7)*T1*TAM(8)*T2*TAM(9)*T3
      T1=XGL(1GL)-XCH
      T2=XGL(1GL)-YCH
      T3=XGL(1GL)-ZCH
      XG(2)=TAM(1)*T1*TAM(2)*T2*TAM(3)*T3
      YG(2)=TAM(4)*T1*TAM(5)*T2*TAM(6)*T3
      ZG(2)=TAM(7)*T1*TAM(8)*T2*TAM(9)*T3
      IF(TMA.GT.0.0) GO TO 1
      XG(1)=-XG(1)
      XG(2)=-XG(2)
      10
      15
      0=XG(1)-XG(2)
      0=YG(1)-YG(2)
      0=ZG(1)-ZG(2)
      IF(OX.NE.0.0) GO TO 16
      IF(XG(1).LT.0.0) GO TO 10
      IF(XG(1).GT.RCHA) GO TO 10
      IF(OZ.NE.0.0) GO TO 2
      IF(OY.EQ.0.0) GO TO 10
      WJ=(XG(1)*THA)**2-ZG(1)**2
      IF(Y3.LT.0.0) GO TO 10
      IF(Y3.LT.0.0) GO TO 10
      Y3=SIGN(SQR((Y3)*DY))
      OFLIT=IVG(1)-Y3/DY
      IF(OFLIT.GT.1.0) GO TO 10
      IF(OELIT.GE.0.0) GO TO 9
      OFLIT=(Y641+Y3)/DY
      GO TO 7
      20
      25
      30
      35
      40
      45
      50
      55
  
```

1 A=0\*\*2\*DZ\*\*2  
 2 TEMP=YG(1)+ZG(2)-YC(2)+ZG(1)  
 3 =-2.0\*TEMP\*DY  
 4 =TEMP\*\*2-(XG(1)\*THA\*WJ)\*\*2  
 5 =0.0-4.0\*A\*C  
 6 =SORT(0)

	77/09/04 • 14.10.44	PAGE	2
1	FTN 4.6+420		
2	CROUTINE CONFUSE	74/74	OPT=1
3	$L_5 = (-B + SIGN(D, D2)) / (2.0 * A)$	SAH	2217
4	$DFLT = (2G(1) - 73)/D2$	SAH	2218
5	IF (DFLT .LT. 0.0) GO TO 3	SAH	2219
6	IF (DFLT .GT. 1.0) GO TO 9	SAH	2220
7	$L_3 = (-B - SIGN(D, D2)) / (2.0 * A)$	SAH	2221
8	$DFLT = (2G(1) - 73)/D2$	SAH	2222
9	GO TO 7	SAH	2223
10	$A = D * (2.0 * D2)**2 - (0.0 * IMA)**2$	SAH	2224
11	$T_1 = YG(2) * YG(1) - YG(1) * YG(2)$	SAH	2225
12	$T_2 = ZG(2) * XG(1) - ZG(1) * XG(2)$	SAH	2226
13	$A = 2.0 * (T_1 * CY + T_2 * DZ)$	SAH	2227
14	$C = T_1 * 2.0 * T_2 * 2$	SAH	2228
15	IF (A .NE. 0.0) GO TO 5	SAH	2229
16	IF (B .GE. 0.0) GO TO 10	SAH	2230
17	$X_3 = -C/B$	SAH	2231
18	IF (X_3 .GT. RCHA) GO TO 10	SAH	2232
19	$DFLT = (XG(1) - X3)/DX$	SAH	2233
20	GO TO 7	SAH	2234
21	$C = B * E - 4.0 * A * C$	SAH	2235
22	IF (D .LT. E .0) GO TO 10	SAH	2236
23	$E = SORT(D)$	SAH	2237
24	$X_3 = 4.0 * SIGN(D, ADX1) / (2.0 * A)$	SAH	2238
25	IF (X_3 .LT. 0.0) GO TO 10	SAH	2239
26	IF (X_3 .GT. RCHA) GO TO 6	SAH	2240
27	$DFLT = (YG(1) - X3)/DX$	SAH	2241
28	IF (DFLT .GE. 0.0) GO TO 6	SAH	2242
29	$X_3 = (-B - SIGN(D, A * DX1) / (2.0 * A))$	SAH	2243
30	IF (X_3 .LT. 0.0) GO TO 10	SAH	2244
31	IF (X_3 .GT. RCHA) GO TO 10	SAH	2245
32	$DFLT = (XG(1) - X3)/DX$	SAH	2246
33	IF (DFLT .LT. 0.0) GO TO 10	SAH	2247
34	IF (DFLT .GT. 1.0) GO TO 10	SAH	2248
35	IF (DFLT .LT. DFLT) DELT = DFLT	SAH	2249
36	CONTINUE	SAH	2250
37	RETURN	SAH	2251
38	END	SAH	2252

SECTION 5

DEFINITION OF FORTRAN VARIABLES

The following pages of this section list and define those FORTRAN variables which have been added to the DSAMAM program for the endgame. Also shown is the common block in which the variable is found or, if it is not in common, the subroutine which uses it. Finally, there is an indication as to whether the value of the variable is read as input or computed internally.

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block or Subroutine Where Found</u>	<u>Definition</u>
ABL(I)	Input	/TWO/	For the Ith blast ellipsoid, the length of the semi-axis in the X-direction of the aircraft body system (ft)
ASH(I)	Input	/TWO/	For the Ith fragment shield ellipsoid, the length of the semi-axis in the X-direction of the aircraft body system (ft)
BBL(I)	Input	/TWO/	For the Ith blast ellipsoid, the length of the semi-axis in the Y-direction of the aircraft body system (ft)
BETAX	Internal	SPRAY	Direction cosine with respect to the X-axis of the aircraft body system of the line traversed by the fragment
BETAY	Internal	SPRAY	Direction cosine with respect to the Y-axis of the aircraft body system of the line traversed by the fragment
BETAZ	Internal	SPRAY	Direction cosine with respect to the Z-axis of the aircraft body system of the line traversed by the fragment
BSH(I)	Input	/TWO/	For the Ith fragment shield ellipsoid, the length of the semi-axis in the Y-direction of the aircraft body system (ft)
CBL(I)	Input	/TWO/	For the Ith blast ellipsoid, the length of the semi-axis in the Z-direction of the aircraft body system (ft)
CDR	Input	/THREE/	Fragment drag coefficient

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block or Subroutine Where Found</u>	<u>Definition</u>
COSGAM	Internal	SPRAY	Cosine of the angle between the missile velocity and the velocity vector of the fragment in a dynamic explosion
COSTHT	Internal	SPRAY	Cosine of the angle between the missile velocity vector and the velocity vector of the fragment in a static explosion
CPS(I,J)	Internal	/SIX/	Cumulative survival probability for component J at kill level I
CPKAC(I)	Internal	MISSIL	Cumulative overall aircraft kill probability at kill level I
CPST(I,J)	Internal	MISSIL	Cumulative survival probability for component J at kill level I adjusted for doubly vulnerable pairs in order to calculate overall aircraft kill probability
CSH(I)	Input	/TWO/	For the Ith fragment shield ellipsoid, the length of the semi-axis in the Z-direction of the aircraft body system (ft)
DELT	Internal	/ONE/	That fraction of the time increment between two integration points at which radar fuzing occurs
DTCF	Input	/THREE/	Delay time of the contact fuze (sec.)
DTRF	Input	/THREE/	Delay time of the radar fuze (sec.)
EL	Internal	SPRAY	The distance traveled by a fragment before hitting an aircraft component (ft)

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block or Subroutine Where Found</u>	<u>Definition</u>
ELDOT	Internal	SPRAY	Rate of change of EL with respect to time (ft/sec).
EX(I)	Input	/TWO/	X-coordinate in the aircraft body system of the center of gravity of the Ith vulnerable component (ft).
F	Internal	SPRAY	A function of time whose roots represent solutions for the intersection of the fragment and an aircraft component.
FPRIME	Internal	SPRAY	The rate of change of F with respect to time.
FRACT(I)	Input	/FIVE/	Fraction by number of all fragments belonging to fragment mass class I.
HAFANG	Input	/THREE/	Half-angle of the radar fuze (degrees).
HEAD1(I)	Internal	MISSIL	Printout heading. I has a maximum value of 3.
HEAD2(I)	Internal	MISSIL	Printout heading. I has a maximum value of 3.
HEAD3(I)	Internal	MISSIL	Printout heading. I has a maximum value of 3.
HEAD4(I)	Internal	MISSIL	Printout heading. I has a maximum value of 3.
HEAD5(I)	Internal	MISSIL	Printout heading. I has a maximum value of 3.
HEAD6(I)	Internal	MISSIL	Printout heading. I has a maximum value of 3.
I	Internal	/ONE/	Index which points to the particular missile in the air being considered. This variable is also used as a temporary index in other parts of the program which do not relate to the endgame.

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
KLEV(I,J)	Internal	MISSIL	Printout heading which designates kill level J. I has a maximum value of 2.
NBLAST	Input	/ONE/	Number of blast ellipsoids.
NCOMPS	Input	/ONE/	Number of vulnerable components (including doubly vulnerable components).
NDHK	Input	/THREE/	Number of direct-hit-with-kill ellipsoids.
NDHNK	Input	/THREE/	Number of direct-hit-no-kill ellipsoids.
NDUBLY	Input	/THREE/	Number of <u>pairs</u> of doubly vulnerable components.
NGLIT	Input	/ONE/	Number of glitter points.
NLEVS	Input	/ONE/	Number of kill levels.
NSHLDs	Input	/ONE/	Number of shield ellipsoids.
NWTS	Input	/ONE/	Number of fragment mass classes.
PDCF	Input	/THREE/	Probability that the contact fuze is a dud.
PDRF	Input	/THREE/	Probability that the radar fuze is a dud.
PKAC(I)	Internal	MISSIL	Single-shot overall aircraft kill probability at kill level I.
PS(I,J)	Internal	MISSIL	Single-shot survival probability for component J at kill level I.
PSCF(I,J)	Internal	MISSIL	Single-shot survival probability for component J at kill level I due to detonation initiated by contact fuze.

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
PSRF(I,J)	Internal	MISSIL	Single-shot survival probability for component J at kill level I due to detonation initiated by radar fuze.
RADFUZE	Internal	MISSIL	Indicator set to .TRUE. if radar fuze is activated - otherwise .FALSE.
RCHA	Internal	/ONE/	Maximum range of the radar fuze multiplied by the cosine of the radar fuze half-angle (ft).
RMISS	Input	/THREE/	Range outside of which the missile is considered to have missed after having passed the point of closest approach to the target (ft).
SDOWN	Internal	SPRAY	The fraction of its initial velocity to which a fragment slows down after traveling a distance EL.
SLOW(I)	Internal	/FIVE/	Slowdown constant for fragments of mass class I ( $\text{ft}^{-1}$ ).
SPREAD	Internal	SPRAY	Factor which when multiplied by the dynamic fragment density gives the fragment density for a static explosion.
STEAR(I)	Input	/FIVE/	Static fragment density (fragments/steradian). The index I represents 19 points from $0^\circ$ to $180^\circ$ off the nose at $10^\circ$ intervals.
SUM(I)	Internal	SPRAY	Expected number of hits capable of producing kill level I on a component.

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
TAM(I)	Internal	/TWO/	Transformation matrix between the aircraft body coordinate system and the missile body coordinate system. The index I runs from 1 to 9 starting at the upper left of the matrix and proceeding left to right across the three rows of the matrix.
THA	Internal	/ONE/	Tangent of the radar fuze half-angle.
THETA	Internal	SPRAY	Angle between the missile velocity vector and the fragment velocity vector in a static explosion (radians).
TIA(I)	Internal	/AIRCF/	Transformation matrix between the aircraft body coordinate system and the inertial coordinate system. The index I runs from 1 to 9 starting at the upper left of the matrix and proceeding left to right across the three rows of the matrix.
TMI(I)	Internal	/TWO/	Transformation matrix between the missile body coordinate system and the inertial coordinate system. The index I runs from 1 to 9 starting at the upper left of the matrix and proceeding left to right across the three rows of the matrix.
VE	Input	/THREE/	Fragment static emission speed (ft/sec).
VHIT	Internal	SPRAY	The speed of the striking fragment (ft/sec).

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
VNET	Internal	SPRAY	The net striking speed of the fragment (i.e., the magnitude of the relative velocity vector between the fragment and the aircraft) (ft/sec).
VNETHI	Input	/ONE/	Normalizing velocity for vulnerable area inputs. Values are read for each of the 11 net striking velocities corresponding to 0, 0.1, 0.2,...1.0 times VNETHI (ft/sec).
VODOT	Internal	SPRAY	Rate of change of VZERO with respect to time (ft/sec <sup>2</sup> ).
VS(I)	Internal	SPRAY	Missile velocity vector in the aircraft body system (ft/sec).
VT(I)	Internal	SPRAY	Aircraft velocity vector in the aircraft body system (ft/sec).
VUL(I,J,K)	Input	/FIVE/	Vulnerable area (ft <sup>2</sup> ). The index I represents kill level. The index J represents 11 levels of net striking speed. The index K combines aspect, fragment mass class, and component.
VX	Internal	MISSIL	X-component in the inertial coordinate system of the relative velocity vector of the aircraft with respect to the missile (ft/sec).

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
VY	Internal	MISSIL	Y-component in the inertial coordinate system of the relative velocity vector of the aircraft with respect to the missile (ft/sec).
VZ	Internal	MISSIL	Z-component in the inertial coordinate system of the relative velocity vector of the aircraft with respect to the missile (ft/sec).
VZERO	Internal	SPRAY	Magnitude of the dynamic velocity vector of the fragment (ft/sec).
WATE(I)	Input	/FOUR/	Average weight of fragments in mass class I (grains).
WY(I)	Input	/TWO/	Y-coordinate in the aircraft body system of the center of gravity of the Ith vulnerable component (ft.).
XBL(I)	Input	/TWO/	X-coordinate in the aircraft body system of the center of the Ith blast ellipsoid (ft.).
XCG	Internal	MISSIL	X-coordinate in the inertial system of the center of gravity of the aircraft relative to the missile (ft.).
XCM	Internal	/ONE/	X-coordinate in the aircraft body system of the missile currently being considered (ft.).
XCMOLD(I)	Internal	/TWO/	X-coordinate in the aircraft body system of missile I at the previous time step (ft.).

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
XG(I)	Internal	CONFUZE	X-coordinate in the missile body system of the glitter point. The index I is 1 for the previous time pulse and 2 for the current time pulse.
XGL(I)	Input	/TWO/	X-coordinate in the aircraft body system of the Ith glitter point (ft.).
XO(I,J)	Input	/FOUR/	X-coordinate in the aircraft body system of the center of the Ith direct-hit ellipsoid of type J (ft.). The type, J, specifies direct-hit-without-kill (J=1) or direct-hit-with-kill (j=2).
XS(I,J)	Input	/FOUR/	For the Ith direct-hit ellipsoid of type J, the length of the semi-axis in the X-direction of the aircraft body system (ft.). The type, J, specifies direct-hit-without-kill (J=1) or direct-hit-with-kill (J=2).
XSH(I)	Input	/TWO/	X-coordinate in the aircraft body system of the center of the Ith fragment shield ellipsoid (ft.).
XSTAR	Internal	SPRAY	X-coordinate in the aircraft body system of the explosion point (ft.).
YBL(I)	Input	/TWO/	Y-coordinate in the aircraft body system of the center of the Ith blast ellipsoid (ft.).

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
YCG	Internal	MISSIL	Y-coordinate in the inertial system of the center of gravity of the aircraft relative to the missile (ft.).
YCM	Internal	/ONE/	Y-coordinate in the aircraft body system of the missile currently being considered (ft.).
YCMOLD(I)	Internal	/TWO/	Y-coordinate in the aircraft body system of missile I at the previous time step (ft.).
YG(I)	Internal	CONFUZE	Y-coordinate in the missile body system of the glitter point. The index I is 1 for the previous time pulse and 2 for the current time pulse.
YGL(I)	Input	/TWO/	Y-coordinate in the aircraft body system of the Ith glitter point (ft.).
YO(I,J)	Input	/FOUR/	Y-coordinate in the aircraft body system of the center of the Ith direct-hit ellipsoid of type J (ft.). The type, J, specifies direct-hit-without-kill (J=1) or direct-hit-with-kill (J=2).
YS(I,J)	Input	/FOUR/	For the Ith direct-hit ellipsoid of type J, the length of the semi-axis in the Y-direction of the aircraft body system (ft.). The type, J, specifies direct-hit-without-kill (J=1) or direct-hit-with-kill (J=2).

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
YSH(I)	Input	/TWO/	Y-coordinate in the aircraft body system of the center of the Ith fragment shield ellipsoid (ft.).
YSTAR	Internal	SPRAY	Y-coordinate in the aircraft body system of the explosion point (ft.).
ZBL(I)	Input	/TWO/	Z-coordinate in the aircraft body system of the center of the Ith blast ellipsoid (ft.).
ZCG	Internal	MISSIL	Z-coordinate in the inertial system of the center of gravity of the aircraft relative to the missile (ft.).
ZCM	Internal	/ONE/	Z-coordinate in the aircraft body system of the missile currently being considered (ft.).
ZCMOLD(I)	Internal	/TWO/	Z-coordinate in the aircraft body system of missile I at the previous time step.
ZEE(I)	Input	/TWO/	Z-coordinate in the aircraft body system of the center of gravity of the Ith vulnerable component (ft.).
ZG(I)	Internal	CONFUZE	Z-coordinate in the missile body system of the glitter point. The index I is 1 for the previous time pulse and 2 for the current time pulse.
ZGL(I)	Input	/TWO/	Z-coordinate in the aircraft body system of the Ith glitter point (ft.).

<u>FORTRAN Variable</u>	<u>Input or Internal</u>	<u>Common Block of Subroutine Where Found</u>	<u>Definition</u>
Z0(I,J)	Input	/FOUR/	Z-coordinate in the aircraft body system of the center of the Ith direct-hit ellipsoid of type J (ft.). The type, J, specifies direct-hit-without-kill (J=1) or direct-hit-with-kill (J=2).
ZS(I,J)	Input	/FOUR/	For the Ith direct-hit ellipsoid of type J, the length of the semi-axis in the Z-direction of the aircraft body system (ft.). The type, J, specifies direct-hit-without-kill (J=1) or direct-hit-with-kill (J=2).
ZSH(I)	Input	/TWO/	Z-coordinate in the aircraft body system of the center of the Ith fragment shield ellipsoid (ft.).
ZSTAR	Internal	SPRAY	Z-coordinate in the aircraft body system of the explosion point (ft.).