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AIRCREW INFLIGHT PHYSIOLOGICAL DATA ACQUISITION SYSTEM II.(U)

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DATA ACQUISITION SYSTEM II

THESIS

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AFIT/GE/EE/77-21

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**AIRCREW INFLIGHT PHYSIOLOGICAL  
DATA ACQUISITION SYSTEM II**

**THESIS**

**Presented to the Faculty of the School of Engineering  
of the Air Force Institute of Technology  
Air University  
in Partial Fulfillment of the  
Requirements for the Degree of  
Master of Science**

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**Graduate Electrical Engineering**

**December 1977**

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## Preface

This thesis is an investigation into a second generation Inflight Physiological Data Acquisition System sponsored by the School of Aerospace Medicine (SAM) at Brooks AFB, Texas. The investigation entailed designing, building, and testing both the hardware and the software for a microprocessor-based prototype system. The resulting prototype, presented in this report, is a complete data acquisition system including the sensor interfaces, the microprocessor, and a permanent memory device. The body of the report is written in general terms for the user at SAM; a more detailed description including circuit diagrams, program listings, and technical discussions is given in the appendices.

This project would not have been possible without the encouragement and help of a number of people. We would like to acknowledge the individuals who helped to make this thesis a reality.

We are indebted to Dr. Mathew Kabrisky, our thesis advisor, who gave us the guidance and enthusiasm to pursue this project. We are grateful to Dr. Gary Lamont, Capt Mike Weber, and Capt Chuck Cornell whose suggestions and assistance helped immeasurably. We would also like to thank Bob Durham and Dan Zambon for their superb technical assistance. We are also grateful to the personnel from the Crew Technology Division at SAM for their support in obtaining the hardware. Our gratitude also goes to Al Haun of Analog Devices for his assistance with the data acquisition module and to Jack Capehart at the ASD computer center for his assistance in transferring the operating system to PROM.

Our deepest gratitude goes to Steve's wife, Cindy, and to Greg's wife, Sue, and son, Jason, for their continuing encouragement and patience while we pursued this project.

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## List of Abbreviations

<u>Abbreviation</u>	<u>Definition</u>
ABSPR	Absolute pressure sensor service routine
A/D	Analog to digital
AFIT	Air Force Institute of Technology
AMRL	Aerospace Medical Research Laboratory
ASCII	American Standard Code for Information Interchange
CPU	Central processing unit
ECG	Electrocardiogram
EOC	End of conversion
EOI	End of interrupt
ERROR	Error correcting routine
FLWRT	Flow rate sensor service routine
FORTRAN	(FORMula TRANslation) Engineering programming language
G's	Acceleration of Gravity
Hz	Hertz (cycles/sec)
ICW	Initialization command word
IFPDAS	Aircrew Inflight Physiological Data Acquisition System
I/O	Input/output
IR	Interrupt request
IS	In-service
LSB	Least significant bit
mm Hg	Millimeters of Mercury

List of Abbreviations

(Continued)

<u>Abbreviation</u>	<u>Definition</u>
MSB	Most significant bit
MUX	Multiplexer
OCW	Operational command word
PO2IN	Inspired oxygen partial pressure sensor service routine
PO2OUT	Expired oxygen partial pressure sensor service routine
PROM	Programmable read-only memory
RAM	Random access memory; read/write memory
ROM	Read-only memory
R-wave	Highest amplitude component of a normal ECG
SAM	School of Aerospace Medicine
SBC	Single board computer
USART	Universal synchronous/asynchronous receiver/transmitter

### Notation

- (name) Signifies positive true logic (e.g., EOC)
- ( $\overline{\text{name}}$ ) Signifies negative true logic (e.g.,  $\overline{\text{EOC}}$ )
- xxxx Signifies a decimal number (e.g., 3184)
- xxxxH Signifies a hexadecimal number (e.g., 0C70H)

Abstract

This paper discusses a second generation microprocessor-based prototype system to acquire, analyze, and store selected environmental and physiological data from a pilot during flight. The Aircrew Inflight Physiological Data Acquisition System (IFPDAS) II consists of an input multiplexer and analog-to-digital converter, a heart rate detector, a microprocessor, and a permanent memory device. The microprocessor's operating system monitors eight sensors, extracts desired information, and stores these reduced data in permanent memory. After the flight, these data are transferred to a land-based computer which completes the data processing and graphs the following environmental and physiological information versus flight time: (1) cabin absolute pressure, (2) cabin altitude, (3) Z-G's, (4) heart rate, (5) breathing rate, (6) minute ventilation volume, (7) inspired oxygen quantity, and (8) expired oxygen quantity.

The completed IFPDAS II prototype provides the desired information well within the required accuracy. It provides the following parameter ranges: (1) heart rate from  $53 \pm .1$  to  $225 \pm 2.2$  b/min, (2) breathing rate from  $4.7 \pm .1$  to  $50 \pm 1$  b/min, (3) minute ventilation volume from 0 to  $100 \pm 2$  l/min, (4) absolute pressure from 0 to  $760 \pm 2$  mm Hg, and (5) G's from -3 to  $+12 \pm .1$  G.

## AIRCREW INFLIGHT PHYSIOLOGICAL DATA ACQUISITION SYSTEM II

### I Introduction

#### Background

The Crew Technology Division of the USAF School of Aerospace Medicine (SAM) at Brooks AFB, Texas, has recognized the need to relate pilot activity to physiological measurements, to apply these relationships to predict aircrew effectiveness, and to formulate equipment design and use criteria to optimize that effectiveness in present and projected flying roles.

Current System. SAM currently has an "Aircrew Inflight Physiological Data Acquisition System" (IFPDAS) which records seven analog functions on cassette tape:

1. A standard time code (to correlate flight events and physiological effects);
2. Pilot voice;
3. ECG;
4. Cabin pressure;
5. Oxygen consumption;
6. Expired flow;
7. Vertical acceleration.

The IFPDAS consists of two subsystems: one to sense and record the data (inflight), the other to reproduce the data (on the ground - after the flight). This data is then converted to digital signals and processed by a digital computer.

There are several problems with the current IFPDAS. It was designed and built using discrete components and is therefore not as reliable as a system based on modern components. Secondly, it doesn't have the capability to acquire triaxial G's, inspired flow volume, or separate input and output oxygen concentrations. Finally, it is highly specialized and, therefore, inflexible without costly design modifications.

Currently, the IFPDAS hardware is being modified by the U.S. Navy. This modification includes two additional analog functions as well as some replacement of discrete components by integrated circuits. The resulting modification should be more reliable; however, it still won't have the complete desired capability and flexibility.

System Standards. Personnel at SAM have projected the design requirements for the second generation system (IFPDAS II), due for production in the early 1980's. IFPDAS II must provide the following primary data:

1. Cabin pressure;
2. Time code;
3. G's (triaxial, if available);
4. Voice (this can be acquired separately if an all-digital system can be designed).

In addition, this new system should provide the capability to assess three or more of the following:

1. Inspired and expired flow;
2. Input and output oxygen concentrations;
3. Heart rate;
4. Mask pressure;
5. Garment pressure;
6. Body core temperature.

The desired range and accuracy for each of the sensors is included in Appendix A.

The IFPDAS II must meet several other system specifications. It must accept probe inputs from 0 to 5 volts which correspond to the appropriate range of each function (for example, 0 - 760 mm Hg for absolute pressure). It should provide a data acquisition time of at least three hours, and it must be time-synchronized to correlate flight events and physiological effects. The IFPDAS was designed to be carried in the pilot's survival vest and, therefore, the original size restrictions must still be met. This means that IFPDAS II must be no larger than 2" x 5" x 9" and must be self-contained, with no external connections to the aircraft. Finally, it is desirable to present a visual display of the status of the device and its probes.

#### Scope of Thesis

The purpose of this investigation was to determine the feasibility of implementing IFPDAS II as a completely digital system to eliminate the mechanical drives and reduce the post-flight computations; and, if digital implementation was practical, to update the existing equipment, increase its reliability, and extend its capabilities.

Feasibility. A search of the current literature revealed that the microprocessor is extending the capabilities of monitoring systems and data acquisition systems in the medical and engineering fields (Ref 1, 2, 3, & 4). A microprocessor controlled system offers several excellent features, the most important of which are flexibility and high reliability. In order to modify the function of a microprocessor-based system, all



that is generally required is a change in the software, often with little or no change in the hardware. In addition, the microprocessor can manipulate the data, extracting the significant information, thereby reducing the amount of permanent storage required. Finally, since the microprocessor incorporates numerous digital functions onto a single unit, or chip, it replaces an enormous number of discrete components, thereby greatly increasing system reliability. For these reasons, the investigation includes the development of a microprocessor-based IFPDAS II prototype.

Assumptions. This IFPDAS II prototype was designed making four assumptions. The first assumption is that the probes supply the desired data. This assumption is required since it is not within the scope of this investigation to redesign the probes. The second assumption is that there is a maximum period of four hours during which data is collected. This is required to establish the memory size needed to store the data, and is justified since the current system is limited to four hours - which personnel at SAM found to be satisfactory. The third assumption is that continuous storage of the data is not required. Instead, a periodic technique (i.e., every 10 to 30 secs) or a "store-on-significant-change" technique could be used. This, too, is necessary to limit the memory size. This assumption should not limit the usefulness of the data since unchanging data is generally not interesting. It is the changes in the data that are important, and both techniques will detect the changes. In addition, it will generally take 10 to 30 seconds to detect changes in, for example, heart rate or flow rate. The fourth and last assumption is that the power and size requirements would not

have to be met for this prototype development. This is required so that proven systems could be used for the development, with less emphasis on their size or power consumption.

System Configuration. The minimum configuration for a microprocessor-based digital data acquisition system would have to include a sensor, an analog-to-digital (A/D) interface, a microprocessor, and a memory device. When more than one sensor is required, the A/D interface becomes more complex. In order to keep this interface to a minimum, the sensor inputs can be multiplexed to one A/D converter, rather than using an A/D converter for each sensor. This not only reduces the number of converters, but reduces the number of inputs to the microprocessor. The multiplexer and single A/D converter configuration allows additional sensors to be interfaced to the system without changing the basic system hardware, while keeping the number of system components to a minimum. These concepts were used to design the IFPDAS II prototype, which consists of six major functional units as shown in Figure 1.

The multiplexer selects one from up to 16 different probe signals as the input to the A/D converter. This converter transforms the analog signal into a digital number which represents the signal for use by the microprocessor (CPU). The heart rate detector (implemented in hardware for required accuracy - see ECG section of Chapter II) supplies the CPU with a number representing the heart beat interval. The CPU combines these numbers with previous data (stored temporarily in the CPU memory), extracts the desired information, and stores the desired result in permanent memory.

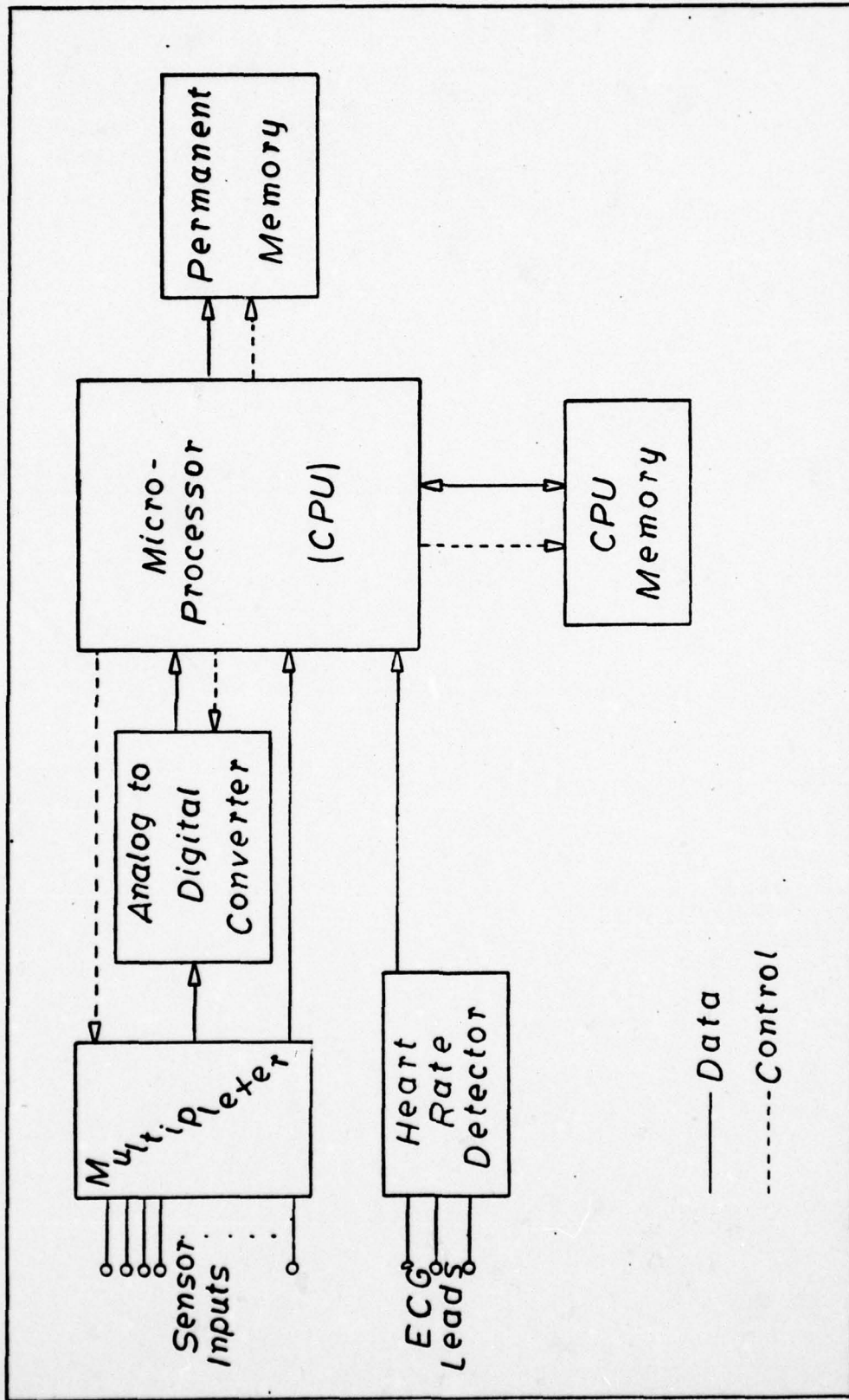


Fig. 1. System Components

The microprocessor performs two major functions: data manipulation and system control. Both of these functions are implemented by a software program, called the operating system, which is stored in the CPU memory. The operating system directs selection of the proper input probe, initiates conversion of the analog signal, and routes data to and from both CPU and permanent memory. The operating system also ensures proper interpretation and reduction of the data.

The remaining chapters discuss the IFPDAS II prototype with respect to its hardware, the general algorithms designed to acquire the data, the post-flight data reduction, and the tests of the prototype system. Recommendations for further system development are also given.

## II Hardware

Several microprocessors (including the Z80, the TMS9900, the 8080, and the 6800) were considered for use as the central processing unit for the IFPDAS II prototype. Intel Corporation's 8080A CPU has several advantages over the other microprocessors which make it the best choice for the system CPU. A major advantage is that the 8080 is a well established, low cost, highly reliable microprocessor (Ref 5:44-45) which readily integrates with Intel's general purpose peripherals. These peripherals provide a variety of special functions (such as timing and external interfaces) which, together with the 8080, make up a complete computer system. In addition, the devices selected for the A/D interface and the permanent memory had already been interfaced to the 8080. Other considerations include the author's previous experience with the 8080 and the support software available on AFIT's computer system. One final advantage is Intel's new generation 8080A - the 8085 - which is 100% software compatible with the 8080 and integrates several functions of the 8080 system onto a single chip (Ref 6:109-113, Ref 7).

The IFPDAS II prototype consists of four major hardware components: Intel's SBC 80/20 (a single board computer containing the 8080 and several peripherals), the DAS 1128 Data Acquisition Module, the sensor interfaces, and the permanent data storage device. Each of these components will be described in the following sections. IFPDAS II prototype characteristics are listed in Appendix A.

### SBC 80/20

Intel's 8080-based single board computer SBC 80/20 was purchased. This allowed engineering development of the IFPDAS II prototype using a proven computer system rather than devoting time to fabrication and testing of a specialized computer system.

The SBC 80/20 is a complete computer system on a single 6.75-by-12 inch printed circuit card. The CPU, system control functions, CPU memories, input/output (I/O) interfaces, interval timers, and interrupt controller all reside on the board (a block diagram of these functions is shown in Figure 2). The CPU functions have been discussed in the section on system configuration; each of the peripheral devices will be introduced in the remainder of this section. (Specific design and operational characteristics are discussed in Appendix E.)

The CPU memory consists of two types of memory: read/write memory (RAM) and read-only memory (ROM). Unlike the read/write memory, the read-only memory is non-volatile, which means that a program or data stored on the ROM will not be lost by turning the power off. For this reason, the operating system will be permanently stored in ROM. Read/write memory is used like a scratch pad by the CPU. Previous data samples, intermediate calculations, and event counters are stored on this temporary memory for later use by the CPU. Any data stored in RAM would be lost if power failed; however, the operating system would recognize the loss of power and reinitialize the system when power is restored.

Communication with the input devices (the data acquisition module and the heart rate detector) is accomplished through two 8255 Programmable Peripheral Interfaces. These 8255s receive data over 8 or 16 data

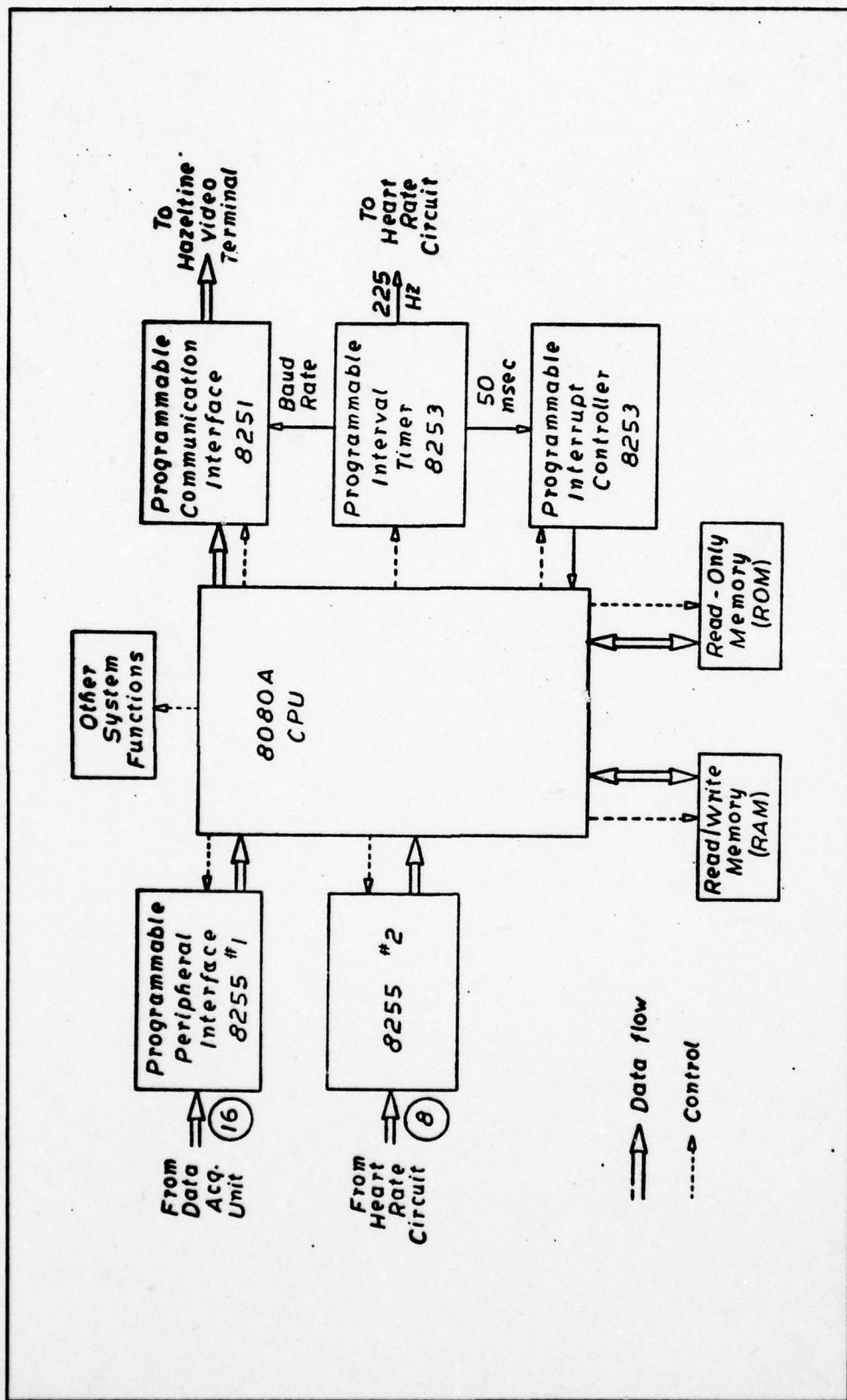


Fig. 2. SBC 80/20 Functional Diagram

lines when informed by an external device that the data is ready for input. The 8255s store the data, indicate to the CPU that the data is available, and transfers the data to the CPU upon request.

Communication with the current memory device (the Hazeltine video terminal) is accomplished through the 8251 Programmable Communication Interface. This device communicates with external devices over a standardized (RS232) interface. (The use of this standardized interface allows communication with any RS232-compatible device - not just the Hazeltine.) The rate of transfer (baud rate or bit rate) and format of the data is controlled by the system software.

Timing and frequency division are accomplished by the 8253 Programmable Interval Timer. The 8253 contains three independent timers, each of which is programmed by the operating system. One of the timers is configured as a real time clock which informs the CPU of every 50 msec interval. The other two timers are used as frequency dividers, slowing the system clock to required frequencies. One of the resulting frequencies is used to establish the baud rate for the 8251 communications interface. The other is used as the clock for the heart rate detector.

The last peripheral device contained on the SBC 80/20 is the 8259 Programmable Interrupt Controller. An interrupt is notification to the CPU by an external device that an event has occurred or that the device requires servicing (Ref 8: Ch 5, 8-33). The 8259 intercepts the interrupt request from the external device, and it informs the CPU that the interrupt has occurred and where the software service routine can be found.



### DAS 1128 Data Acquisition Module

The DAS 1128 is a self-contained data acquisition system manufactured by Analog Devices. This compact module was selected for use in the prototype since it is a proven system which readily integrates with the 8080 through the 8255. It contains an analog input multiplexer, an A/D converter, and all of the timing and control circuitry needed to perform the complete data acquisition function (Figure 3).

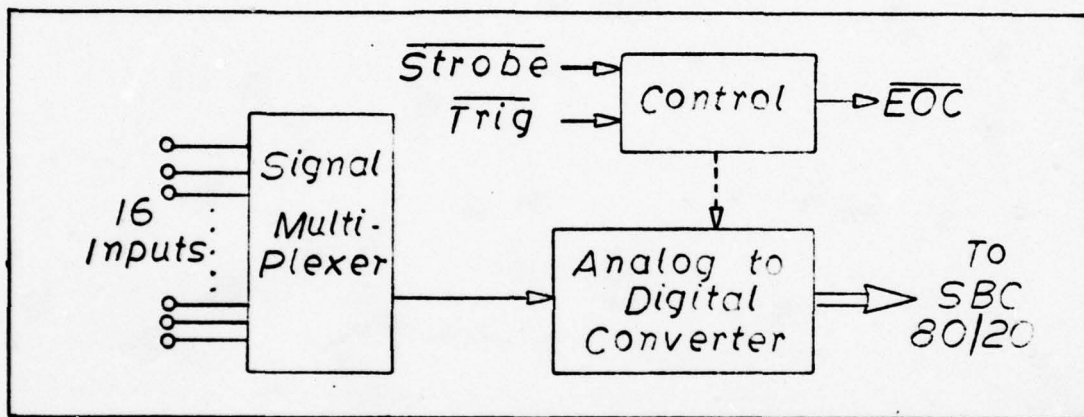


Fig. 3. DAS 1128 Functional Diagram

Seven physiological and environmental analog signals are input to the signal multiplexer. The multiplexer is directed by the CPU (via the  $\overline{\text{STROBE}}$  input) to select these signals sequentially. Conversion of the selected signal is started when the module is triggered by the CPU (via the  $\overline{\text{TRIG}}$  input). (These CPU commands are transmitted by the 8255 to the DAS 1128.) When the conversion is complete, the output of the A/D converter is a digital representation of the 0 - 5 volt input, accurate to within 10 mv of the input signal. (This provides the required

accuracy as listed in Appendix A.) At this time, an "end-of-conversion" (EOC) is returned to the 8255, signifying that the digital data and associated sensor identification are ready for transmission. Each conversion requires approximately 25 microseconds, which would allow a maximum of 40,000 conversions per second. This is more than adequate since the current application requires only 1120 conversions per second. (This is based on a sampling rate of 20 Hz. If a faster rate is desired, the IFPDAS II could sample the maximum 16 inputs at over 70 Hz.) The electrical and interface configurations are discussed in Appendix F.

#### Sensor Interfaces

The IFPDAS II prototype accepts eight physiological and environmental sensors. These probes measure partial pressure of oxygen inhaled and exhaled, respiratory flow rate (expired), triaxial G's, absolute pressure, and ECG. A mask and two oxygen partial pressure sensors (Beckman OM11) were provided by SAM at Brooks AFB, TX. An accelerometer (Statham F-15-340) was obtained from the Aerospace Medical Research Laboratory (AMRL) at Wright-Patterson AFB, OH; and the ECG leads were available in AFIT's Bioengineering Laboratory. (The flow rate and absolute pressure sensors were not available.) Each of the available probes was interfaced to the prototype and is discussed in the following sections.

Oxygen Partial Pressure. The oxygen sensors are located in the mask/hose assembly in the same configuration used for the original IFPDAS (Ref 9:24-25); however, the oxygen information has changed. IFPDAS I provided a measurement representing the instantaneous difference

between the inspired and expired oxygen partial pressures. This is not the information desired; therefore, IFPDAS II provides two measurements, one representing the inspired and one representing the expired oxygen partial pressure. The OM11 sensor output is amplified to provide a 0 - 5 volt signal corresponding to a pressure range of 0 - 760 mm Hg of oxygen.

The OM11 sensor has an 800 msec response time for a 0 to 100% oxygen transition. This is an acceptable time for the present application; however, it is too slow for a breath-by-breath analysis. A complete circuit description of the amplifier is included in Appendix G. Included also is a circuit to reduce the response time to 100 msec, which would be sufficient for a breath-by-breath analysis, if desired.

Acceleration. The IFPDAS II prototype was designed to monitor the acceleration forces in all three dimensions. AMRL at Wright-Patterson AFB has mounted three uni-directional accelerometers on a breast plate (Ref 10) to provide the triaxial G's. Only one sensor was available, so only one amplifier was constructed and tested. However, the other two amplifiers (required for full triaxial G measurement) would be identical to the one actually built. The full range of the Statham accelerometer is -15 to +15 G's. The amplification circuit offsets this range and provides a 0 - 5 volt signal which corresponds to -3 to +12 G's. A complete circuit description of the accelerometer interface is given in Appendix G.

ECG (Heart Rate). IFPDAS I stores the complete ECG on cassette tape for later analysis. This recording is only reliable enough to provide information for heart rate calculation, by hand, on the ground.

Normally, the heart rate is the desired information; therefore a heart rate detector was designed and built for the IFPDAS II prototype. This detector provides a digital representation of the heart rate, along with the provision to record the entire ECG waveform with a separate analog recorder, if desired. The heart rate information is derived by counting the time interval between detected R-waves. This count is passed to the CPU, through the second 8255. A functional diagram of the heart rate detector is given in Figure 4.

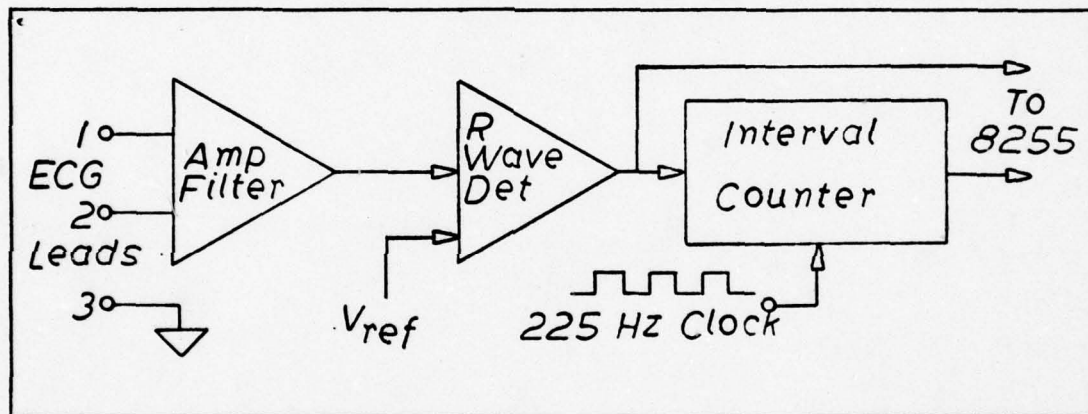


Fig. 4. Heart Rate Detector Functional Diagram

This heart rate detector was designed to provide a maximum amplitude R-wave signal, while eliminating base-line shifts and reducing muscle artifacts. The base-line shifts and muscle artifacts are undesirable since they add extraneous signals and make accurate R-wave detection very difficult. The lead placement shown in Figure 5 is a compromise placement which provides a good QRS-wave least disturbed by muscle artifacts.

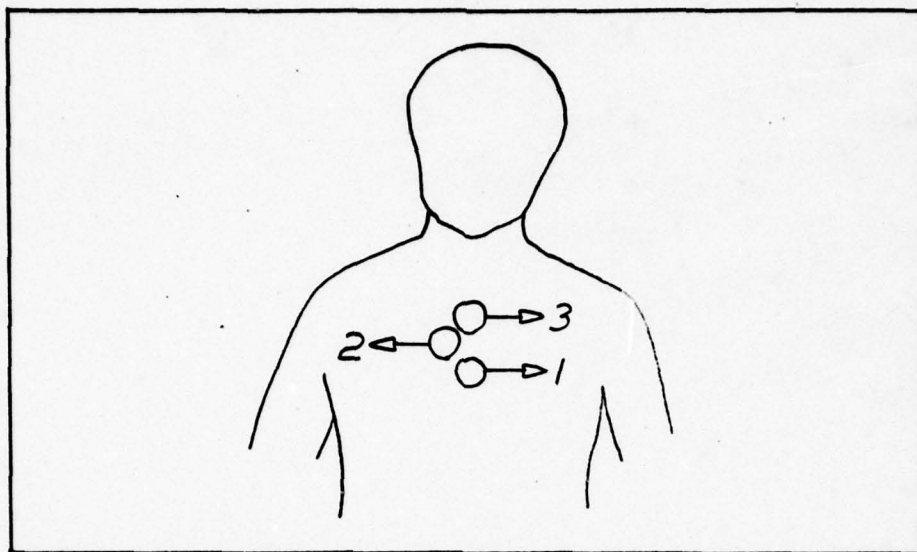


Fig. 5. ECG Lead Placement

Figure 6 shows typical ECG's provided by the amplifier, along with the associated R-wave detector output. Figure 6a was recorded with the subject at rest, while Figure 6b was recorded with the subject exercising heavily on a stationary bicycle. As can be seen from these figures, the R-wave detector provides a highly reliable R-wave indication even under exercise and movement conditions more severe than a pilot would experience during a flight.

The output of the R-wave detector is used to trigger a counter, which counts the time interval between R-waves. When a subsequent R-wave is detected, this count is passed to the 8255 and the new count is started. Appendix G contains a complete electrical description of the ECG amplifier, R-wave detector, and interval counter.

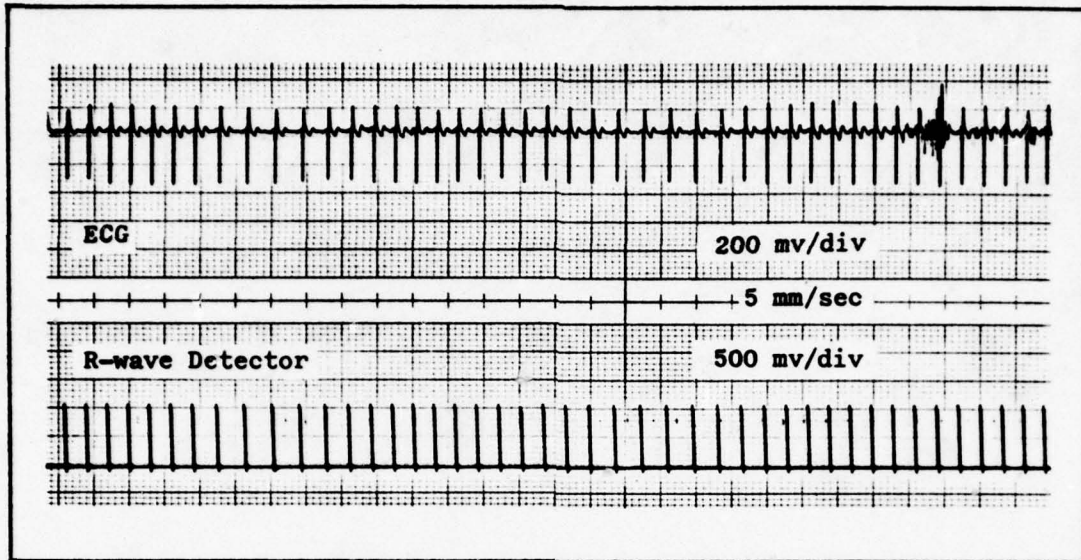


Fig. 6a. ECG & R-wave Detector Output (Rest)

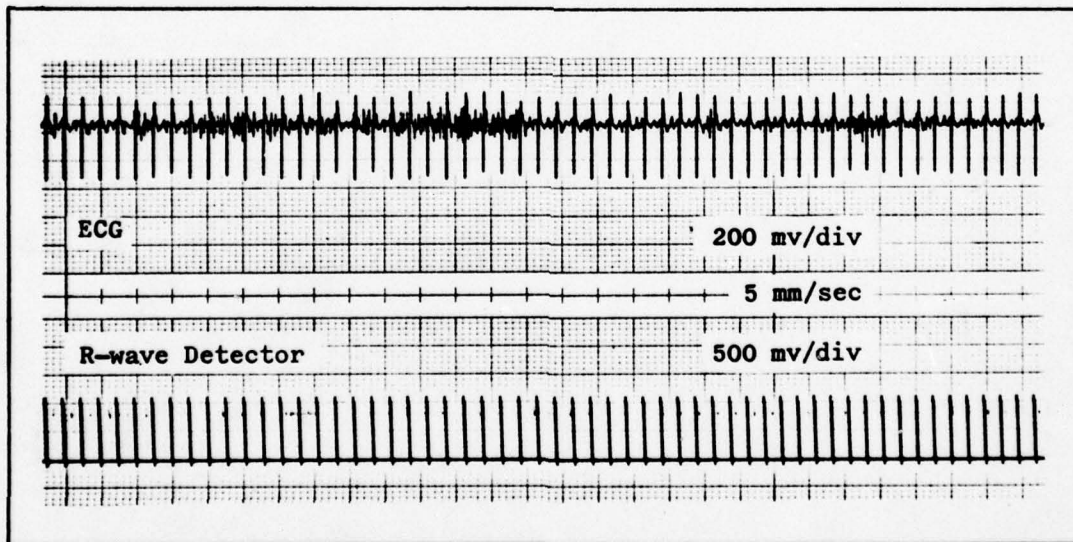


Fig. 6b. ECG & R-wave Detector Output (Exercise)

### Data Storage

A variety of current data storage devices were considered. Several of these were eliminated because of their physical size (magnetic core memory), volatility (RAM, charge-coupled devices), or impracticality (PROM). The only devices considered practical for use on the IFPDAS II prototype were magnetic tape (either wafer or standard cassette) and magnetic bubble memory. (Product information for the two tape storage systems is given in Ref 11, 12.) Both of these storage devices provide non-volatile storage, large capacity, and small size. The magnetic bubble memory was selected over the cassette tape systems in order to provide a completely digital system with no mechanical drives.

Bubble Memory. Bubble memories have several characteristics which make them ideal for use as a mass memory system in this application.

These attributes result primarily from the semiconductor-like processing used to fabricate bubble devices in conjunction with their non-volatile nature. ... Some of the important features resulting from the semiconductor-like processing include reliability, small size, fast access time (relative to electro-mechanical mass storage devices), low cost, and small incremental capacity. ... By virtue of its non-volatility, the bubble device offers removability, asynchronous access, and low power. (Ref 13:1)

Bubble memory systems are still in the developmental stage. It is expected that a complete memory system, including two bubble memories (capable of storing 184,000 bits of information) and all of the system controls and interfacing, will fit on a single 4.5" x 6" x .75" printed circuit board. It is also expected that the power required to operate the system will be approximately three watts. The IFPDAS II prototype only stores data for 10 msec every 10 sec which would allow the CPU

to turn off the memory system during periods of non-storage - thus significantly reducing the power consumption.

Texas Instruments supplied circuit diagrams for the bubble memory controller and functional timing generator (Appendix H). These circuits were wired on a component board and interfaced to the SBC 80/20. Due to limited bubble memory production and procurement difficulties, a complete bubble memory system could not be obtained in time to interface to the IFPDAS II prototype. For this reason a temporary storage method consisting of a Hazeltine 2000 video terminal and digital cassette recorder was used.

Hazeltine Video Terminal. The Hazeltine video terminal communicates with the 8251 Programmable Communication Interface discussed in the SBC 80/20 section of this chapter. The operating system transforms the data into a standard format (seven-bit ASCII code) which is sent to the video terminal and displayed on the video screen. This visual display was a valuable developmental tool because it allowed easy interpretation of the data for validating the software operations as well as checking the accuracy of the stored data. This temporary "storage" technique was used until the system was operating as desired. Then data was collected by displaying it on the video screen and automatically copying it permanently on a digital cassette tape. (Appendix C describes the transfer process.) This digital tape represents the final storage of the "inflight" data, which is ready for "post-flight" conversion and graphic display.



The IFPDAS II prototype hardware configuration includes the SBC 80/20 printed circuit board; and a component board containing the DAS 1128, the heart rate detector, and the bubble memory function timing generator. The analog interfaces that provide the 0 - 5 volt signals are packaged in a 3" x 4" x 5" chassis.

### III General Algorithms

#### Introduction

The operation of the hardware discussed in the previous chapter is under the control of a software program called the operating system (Appendix B). Execution of the operating system by the CPU results in initialization of the system and acquisition, reduction, and storage of the data. The operating system selects the proper input sensor, starts the A/D conversion, and inputs the data for reduction and permanent storage. In addition, it ensures that the proper subprogram, or service routine, receives the data.

The biggest constraint on the IFPDAS II is the amount of available permanent storage for the acquired data. Because of this limitation, it is not possible to store the complete waveforms, or even continuous samples of these waveforms, for the desired four hour period. For this reason, a periodic technique (storing the data every ten sec) was used.

Three of the service routines must sum the input data for each ten second interval. In order to constrain the ten second sum to 16 bits of storage, it was necessary to sample no more than 256 times during the 10 second period. A sampling rate of 20 Hz (200 samples in 10 sec) was selected because the input signals change very slowly (less than 5 Hz).

Data are collected from each input signal every 50 msec using a noise-reducing digital filtering technique. This technique consists of taking 8 consecutive samples of the waveform in .5 msec and averaging these 8 samples to produce the 50 msec reading.

In order to collect data from each input signal at the 20 Hz rate, the operating system sets a hardware timer which interrupts the CPU at the end of 50 msec. This interrupt informs the CPU to start the service routine sequence.

The remainder of the chapter discusses the sensor service routines and the additional system support software.

### Service Routines

There is an independent service routine for each input probe since different information is required from each sensor. Each service routine accepts the 50 msec reading from the averaging routine and performs the required operations to prepare the data for permanent storage.

The following sections discuss the information to be extracted from the input waveforms and the algorithm used in each service routine to reduce the data to the desired form.

P02IN. The personnel at SAM need to assess the pilot's oxygen consumption during flight. One of the required quantities needed to compute this consumption is the amount of oxygen inspired during each breath. To accurately compute the quantity of oxygen inspired, the oxygen partial pressure and the inspired flow rate are required. Currently, the mask/hose assembly does not contain a sensor to measure inspired flow; therefore, a program to approximate the quantity of inspired oxygen was written.

The P02IN service routine was written to sum all of the inspired partial pressure readings for the ten second period and store the sum for averaging after the flight. This sum is then used in conjunction

with the expired flow rate (assuming equal inspired and expired rates) to compute the oxygen intake. (Appendix I discusses a single breath analysis method for computing oxygen uptake.)

P02OUT. The second required quantity needed to compute the pilot's oxygen consumption is the amount of carbon dioxide produced during each breath. Current carbon dioxide sensors do not lend themselves to inflight applications due to their size and weight; therefore, an approximation of the expired oxygen quantity had to be made.

The P02OUT service routine sums the expired partial pressure readings for the ten second period and stores the sum for the post-flight conversion, which is similar to the P02IN methods. (Even though the P02IN and P02OUT service routines are inaccurate in their analysis of the oxygen quantities, the sensor interfaces and data handlers were successfully exercised and produce approximate values. A single breath analysis method for computing the expired oxygen quantity is included in Appendix I.)

Flow Rate (FLWRT). The third quantity required to compute the oxygen consumption is flow volume. A probe in the oxygen mask/hose assembly measures the expired volume flow rate, which can be integrated with respect to time to obtain flow volume.

The integral of the flow rate waveform can be computed by using a rectangular approximation technique, since there is very little change to the flow rate signal during a 50 msec interval. The FLWRT service routine sums and stores the 200 flow rate readings for subsequent multiplication (after the flight) by the known 50 msec interval to produce the resulting integral.

Absolute Pressure (ABS<sub>PR</sub>). The final quantity required to compute the oxygen consumption is the cabin absolute pressure. This is necessary in order to compensate the flow rate reading for altitude and also to compute percent oxygen in the inspired and expired air. The ABS<sub>PR</sub> service routine stores a representation of the cabin absolute pressure every ten seconds.

Breathing Rate (BRRT). The BRRT service routine uses the flow rate sample for breath detection. The routine determines when the exhaled breath has stopped and marks this event as the start of a new breath. The interval between breaths is counted and stored.

X, Y, & Z G's. These service routines search for the maximum and minimum acceleration in each direction during the ten second interval. In order to reduce the effects of transient G impulses, each routine averages eight readings before comparing to the previous minimum and maximum values.

Heart Rate (HEART). The HEART service routine checks to see if the heart rate circuit has input a new heart beat interval. If a new count is available, it is read in and stored. Eight heart beat counts are averaged together to provide a representative heart rate.

#### System Support Software

In order to form the complete operating system, several additional routines are necessary. These routines initialize the IFPDAS II prototype, correct detectable errors, store the data on the permanent storage device, and provide the additional support required by the service routines.

Power-up Routine. This routine is executed when the hardware detects that power has been applied to the IFPDAS II prototype. This program configures the IFPDAS to its data acquisition function. This includes programming the 8251 communication interface, the 8255 input ports, the 8253 timers, and the 8259 interrupt controller as described in the hardware chapter. The DAS 1128 is initialized so that probe "0" is the first to be sampled when the service routine series is started. The final tasks accomplished by the power-up routine are the initialization of the scratch pad storage and the initiation of the service routine series (program loop).

Error Routine. The operating system must ensure that the executing service routine is receiving data from its associated probe. Each service routine accomplishes this by checking the sensor identification information. If the service routine and sensor are mismatched, the operating system must select the proper sensor and return to the program loop. This is accomplished by the error routine.

Permanent Data Storage Routine. The operating system keeps track of the running time, and schedules the storage routine every ten seconds. The storage routine transfers a timing preamble and the data computed by the service routines from the scratch pad storage to the permanent storage device. The data is converted to the ASCII code required by the Hazeltine video terminal, and is transmitted over the RS232 interface. After all of the data is transferred, two control characters are transmitted which directs the Hazeltine system to write the data from the video screen onto the digital cassette tape. Finally, the scratch pad storage area is reinitialized. The data storage routine is the only

device-dependent program in the operating system and is, therefore, the only module that will require modification when the new memory system is interfaced.

All of the service routines and system support software are stored on a single 2708 Erasable and Electrically Programmable Read Only Memory chip. This allows program modification by simply erasing and reprogramming the memory.

#### IV Post-Flight Data Conversion

The post-flight conversions are accomplished by a land-based computer after the flight. The data from the memory device is transferred to the computer where a program completes the data conversion and displays the desired information in graphic form. The form of this display (list, tabular, plot versus time, etc.) can be varied by changing the program.

This investigation utilized AFIT's computer system as the land-based computer. The data from the digital cassette tape is transferred to the computer (Appendix C) for the "post-flight" conversions. A FORTRAN program (Appendix D) reads in this data, converts it back to basic form, completes the calculation of the desired information, and displays this information in graphic form. Three environmental and five physiological parameters are graphed versus time by this conversion program. These are absolute pressure, cabin altitude, Z-G's, heart rate, breathing rate, minute ventilation volume, and inspired and expired oxygen volumes.

The following sections describe these calculations and include a representative graph of the parameter.

##### Absolute Pressure

The absolute pressure data is a number from 0 to 250 which is directly proportional to a pressure range of 0 to 760 mm Hg. The data



is converted to actual cabin absolute pressure using the following formula:

$$\text{abs pressure (mm Hg)} = \frac{\text{(data)}}{250} \times 760 \quad (1)$$

The cabin absolute pressure is plotted versus time, as in Figure 7.

#### Cabin Altitude

The absolute pressure data is also used to compute cabin altitude. There is not a simple mathematical relationship between data in mm Hg and altitude in feet (Ref 14:587); therefore, an approximate relationship was derived using linear regression techniques. A logarithmic curve of the form

$$y = a + b \ln(x) \quad (2)$$

was found to provide the best fit. For the altitude range of 0 to 25,000 feet, the following equation was used:

$$\text{altitude (feet)} = 170,156 - 25,685 \ln(\text{abs pressure (mm Hg)}) \quad (3)$$

This equation is accurate to within 275 feet for the 0 to 25,000 foot range. A sample cabin altitude versus time plot is shown in Figure 8.

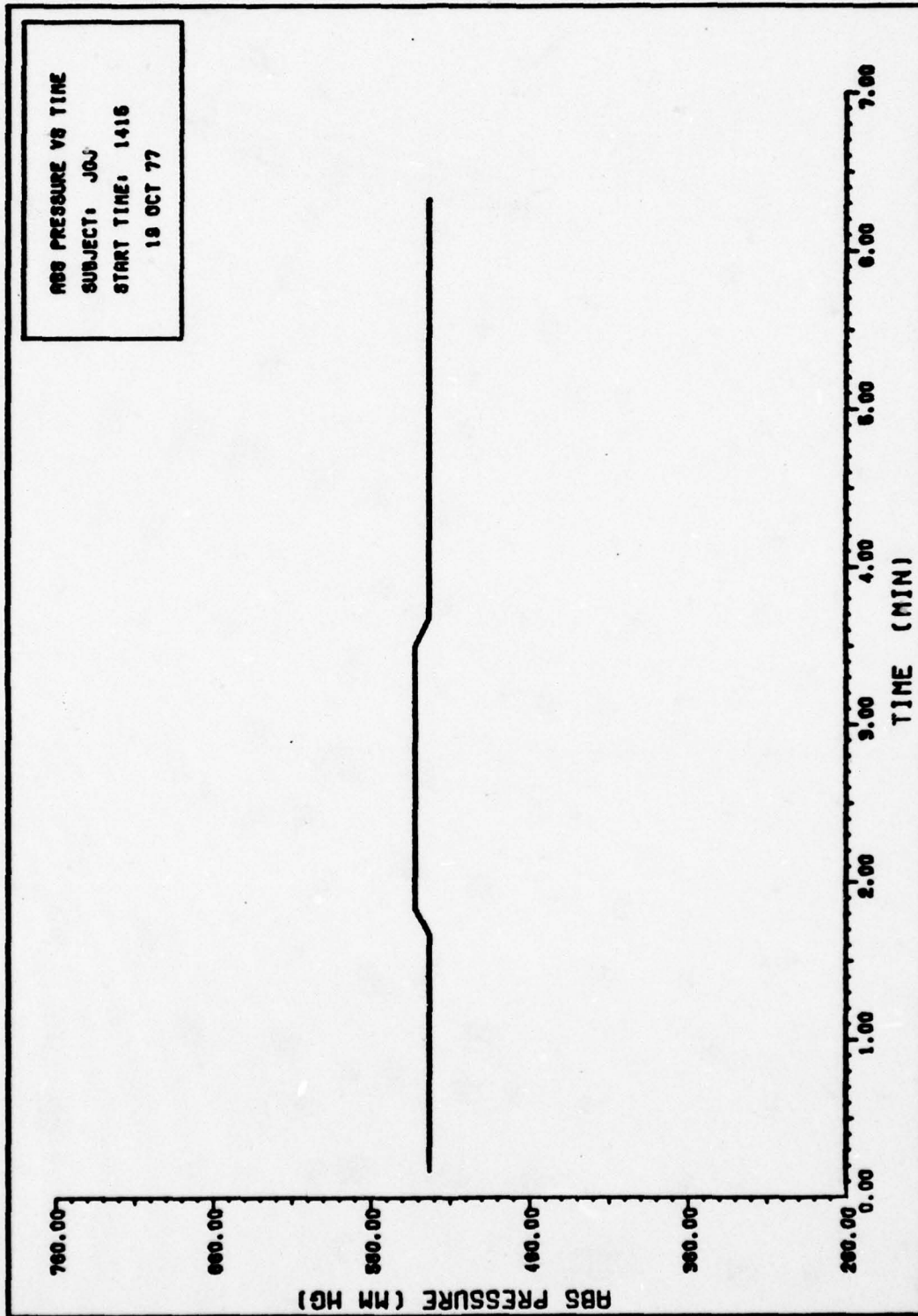


Fig. 7. Cabin Absolute Pressure vs Time

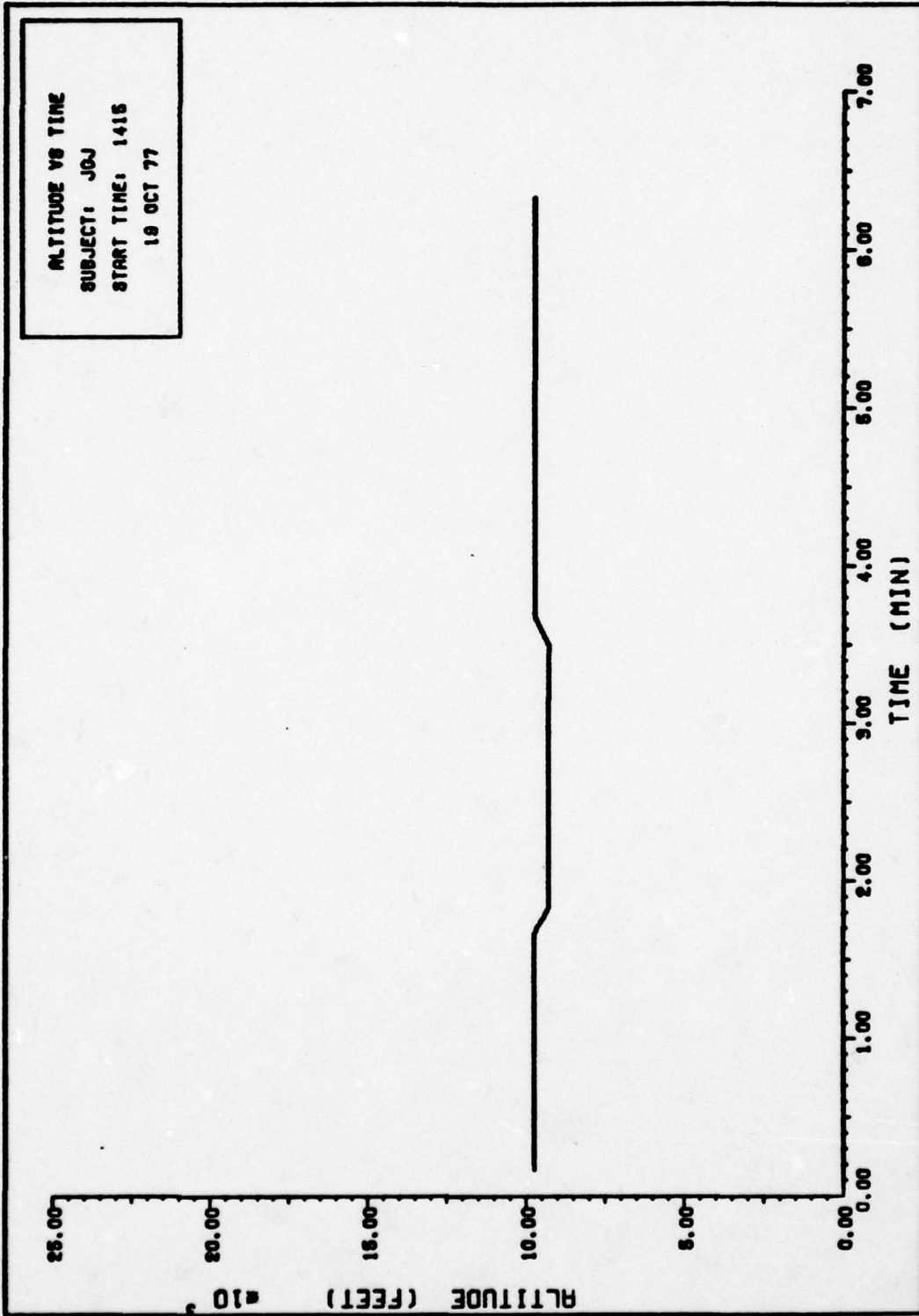


Fig. 8. Cabin Altitude vs Time

### Z-G's

The acceleration data is a number from 0 to 250 which is directly proportional to a range of -3 to +12 G's. The data is converted to actual G's using the following formula:

$$\text{acceleration (G's)} = \left\{ \frac{(\text{data})}{250} \times 15 \right\} - 3 \quad (4)$$

(The -3 term is necessary to correct for the 3 G offset of the acceleration circuitry.) Minimum and maximum Z-G's (for each ten second interval) are plotted on the same graph, as in Figure 9. (The graphs for X and Y-G's are similar and are not included.)

### Heart Rate

The heart rate data is an average number of 4.44 msec (1/225 Hz) counts between detected R-waves. This number is converted to the heart rate using the following equation:

$$\begin{aligned} \text{heart rate (beats/min)} &= \frac{1}{(4.44 \text{ msec})(\text{count})} \times 60 \\ &= \frac{225 \text{ Hz}}{(\text{count})} \times 60 \end{aligned} \quad (5)$$

An example of the heart rate plot is shown in Figure 10.

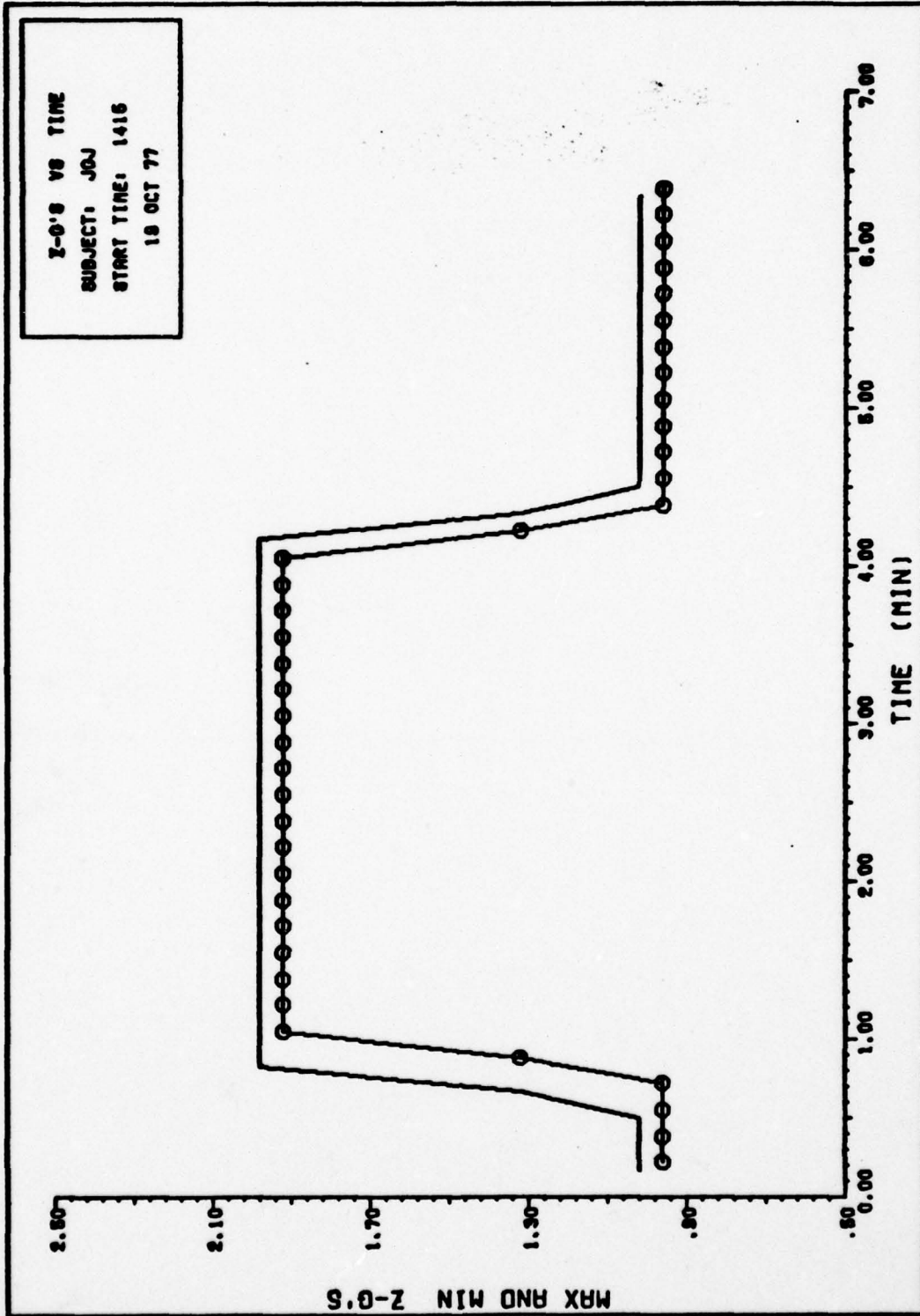


Fig. 9. Minimum and Maximum Z-G's vs Time

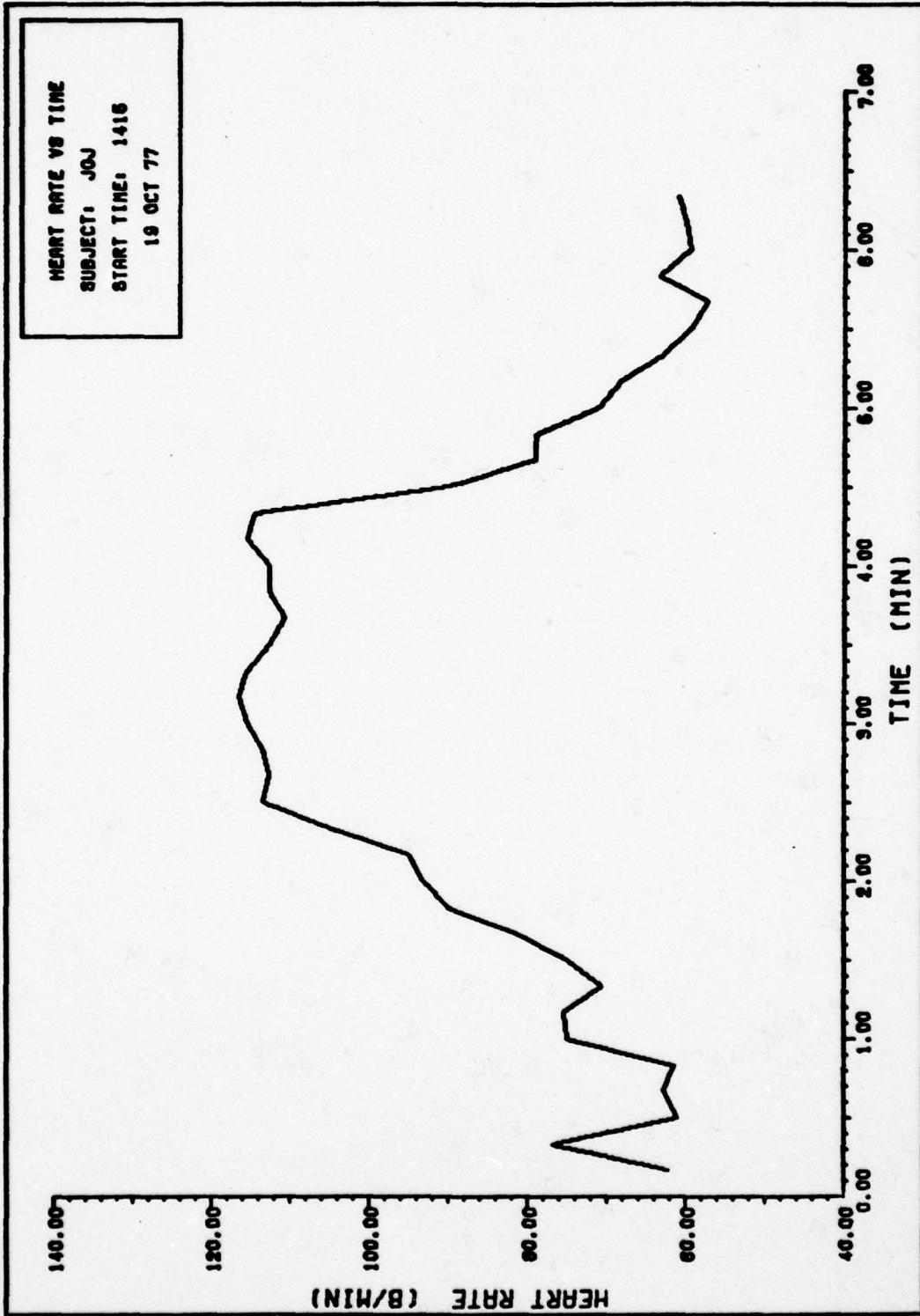


Fig. 10. Heart Rate vs Time

### Breathing Rate

The breathing rate data is similar to the heart rate data, except that each count represents 50 msec. The breathing rate is calculated from the number of 50 msec counts in the following manner:

$$\begin{aligned} \text{breathing rate (breaths/min)} &= \frac{1}{(50 \text{ msec})(\text{count})} \times 60 \\ &= \frac{20 \text{ Hz}}{(\text{count})} \times 60 \end{aligned} \quad (6)$$

The sample breathing rate graph is given in Figure 11.

### Minute Ventilation Volume

The flow volume equations were derived for the original IFPDAS (Ref 9:5). The equation for flow volume rate (F) is

$$F = V_f (\text{volts}) \times (24.8 \text{ l/min/volt}) \times \sqrt{\frac{760 \text{ mm}}{\text{abs pressure}}} \quad (7)$$

(The square root term corrects the flow volume rate for actual altitude.) The 50 msec volume flow rate reading (R) is a number from 0 to 250 which is directly proportional to a flow rate from 0 to 124 l/min. Each R could be used to compute an incremental flow volume expired during that 50 msec interval:

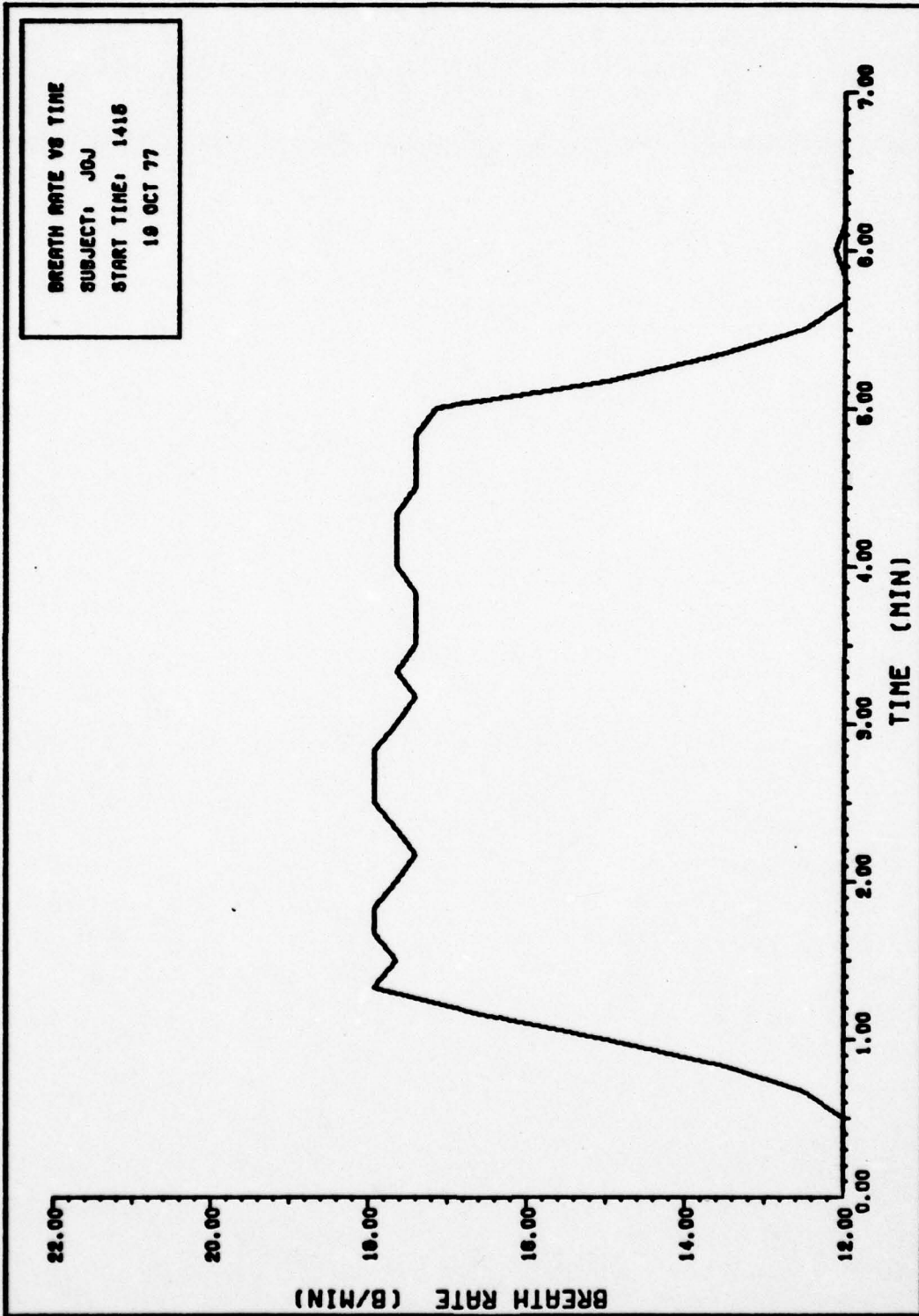


Fig. 11. Breathing Rate vs Time



$$\text{volume/50 msec (1)} = \frac{R}{250} \times (124 \text{ l/min}) \times \sqrt{\frac{760 \text{ mm Hg}}{\text{abs pressure}}} \times \frac{.05 \text{ sec}}{60 \text{ sec/min}} \quad (8)$$

Summing each of the incremental volumes provides the flow volume for the ten second interval:

$$\text{volume/10 sec (1)} = \sum_{i=1}^{200} \left\{ \frac{R_i}{250} \times (124 \text{ l/min}) \times \sqrt{\frac{760 \text{ mm Hg}}{\text{abs pressure}}} \times \frac{.05 \text{ sec}}{60 \text{ sec/min}} \right\} \quad (9)$$

or,

$$\text{volume/10 sec (1)} = \left\{ \frac{124 \text{ l/min}}{250} \times \sqrt{\frac{760 \text{ mm Hg}}{\text{abs pressure}}} \times \frac{.05 \text{ sec}}{60 \text{ sec/min}} \right\} \times \sum_{i=1}^{200} R_i \quad (10)$$

The summation term is the value calculated and stored by the IFPDAS II prototype. Combining constants, Eq (10) reduces to

$$\text{volume}/10 \text{ sec (1)} = .0114 \times \frac{\text{(summation)}}{\sqrt{\text{abs pressure}}} \quad (11)$$

In order to obtain the desired minute ventilation volume, six 10-second volumes are summed. Minute ventilation volume is plotted versus time, as in Figure 12.

### Oxygen Volumes

The inspired and expired oxygen partial pressure data is a sum of the 50 msec readings. This sum is divided by the number of readings, which gives an average oxygen partial pressure reading for the ten second period:

$$\text{avg reading} = \frac{\text{sum}}{200} \quad (12)$$

This average reading is a number between 0 and 250 which is directly proportional to a pressure range of 0 to 760 mm Hg. This data is converted to the fractional amount of oxygen in the air as follows:

$$\text{fraction } O_2 = \left\{ \frac{\text{avg reading}}{250} \times 760 \right\} / \text{abs pressure} \quad (13)$$

The quantity of oxygen in the inspired or expired air is then computed by

$$\text{quantity of } O_2 (1) = (\text{fraction } O_2)(\text{minute vent vol (1)}) \quad (14)$$

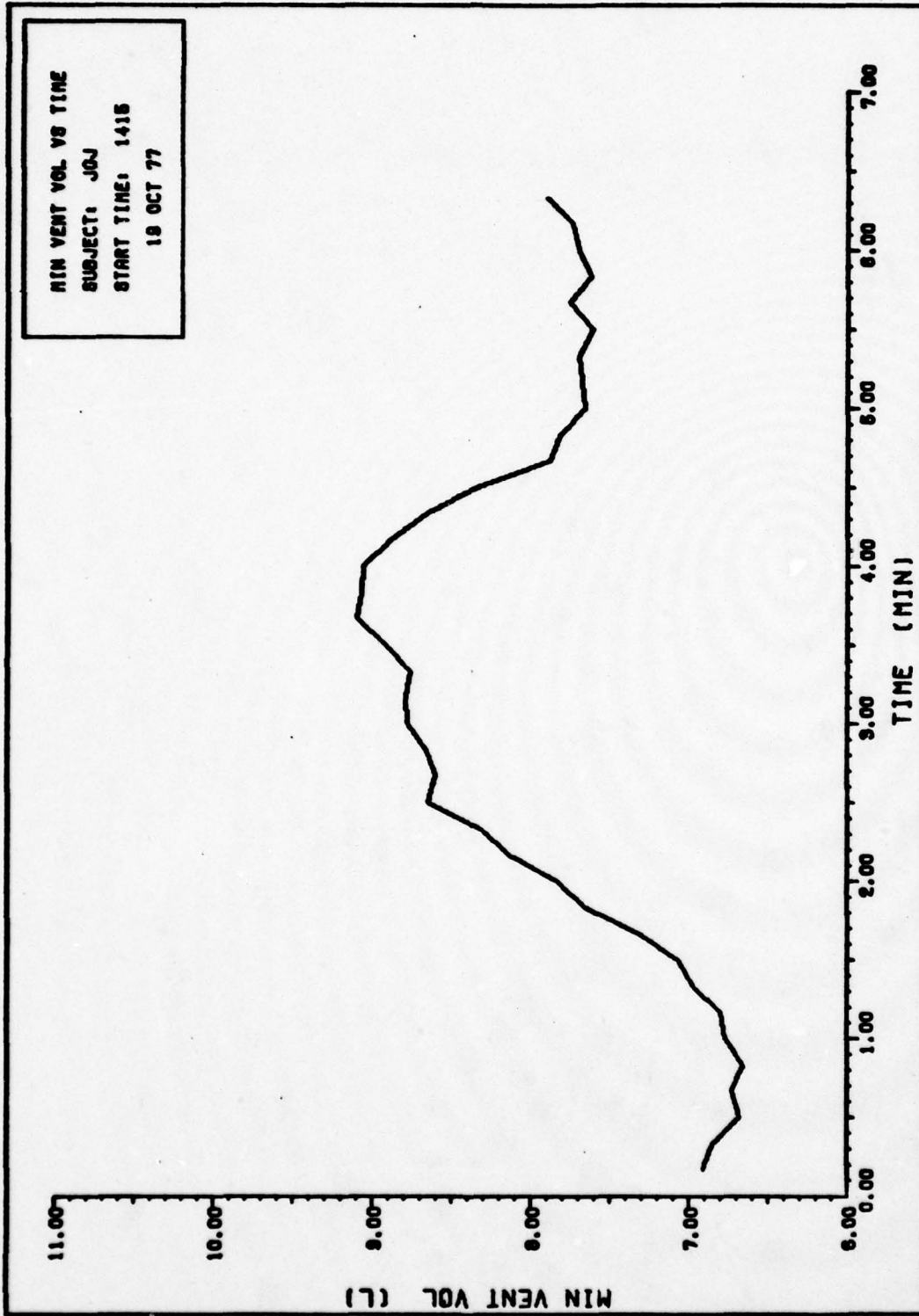


Fig. 12. Minute Ventilation Volume vs Time

Combining Eqs (12), (13), and (14)

$$\text{quantity of } O_2 (l) = \frac{\text{sum X 760}}{200 \text{ X } 250 \text{ X (abs pressure)}} (\text{min vent vol}) \quad (15)$$

or,

$$\text{quantity of } O_2 (l) = \frac{(\text{sum}) \text{ X (min vent vol)}}{65.79 \text{ X (abs pressure)}} \quad (16)$$

The inspired oxygen quantity is graphed in Figure 13; and the expired oxygen quantity is graphed in Figure 14.

The graphs provide a means for easy correlation of the aircraft's pressure altitude and acceleration, and the pilot's heart rate, breathing rate, minute ventilation volume, and oxygen consumption.

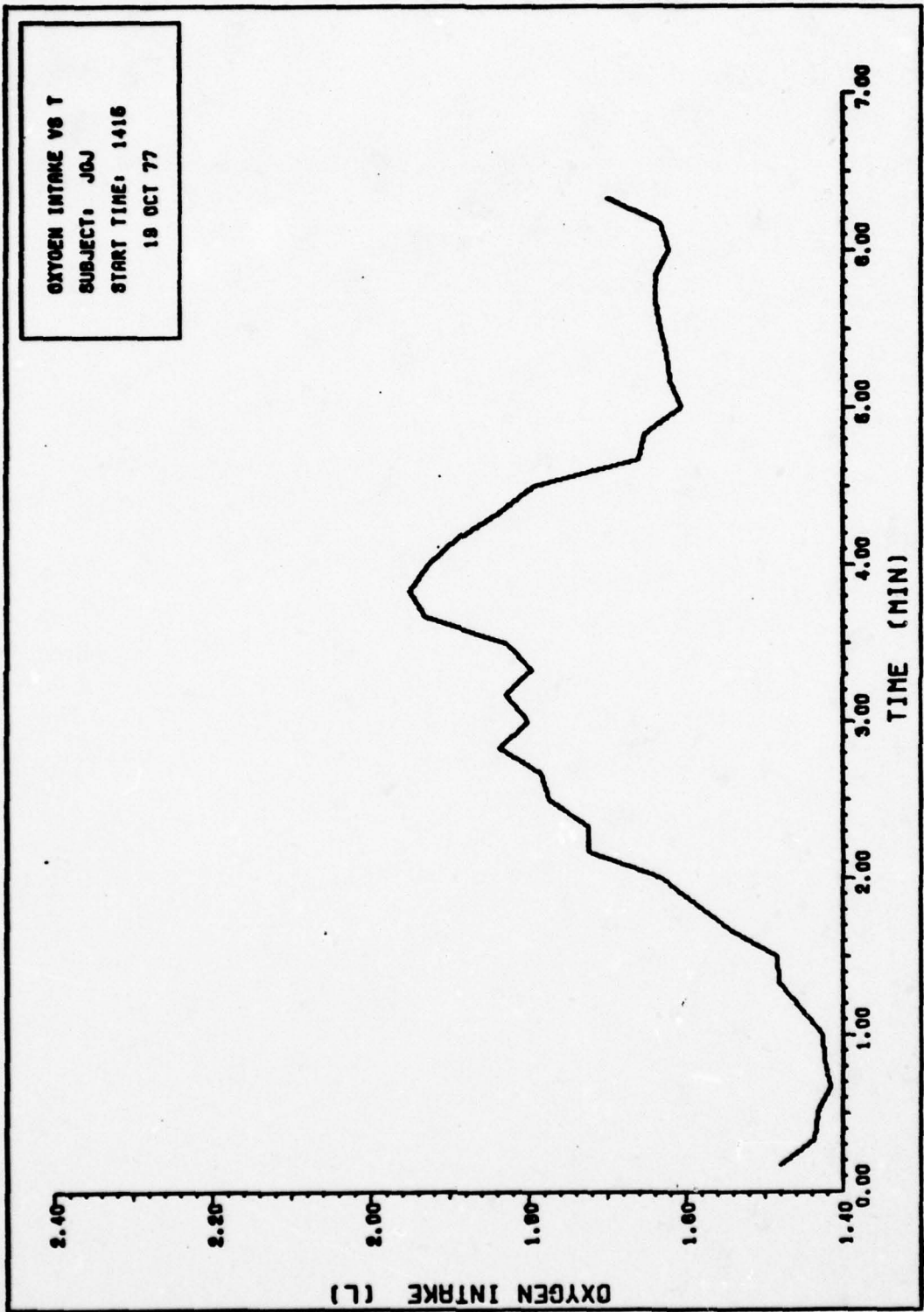


Fig. 13. Inspired Oxygen Quantity vs Time

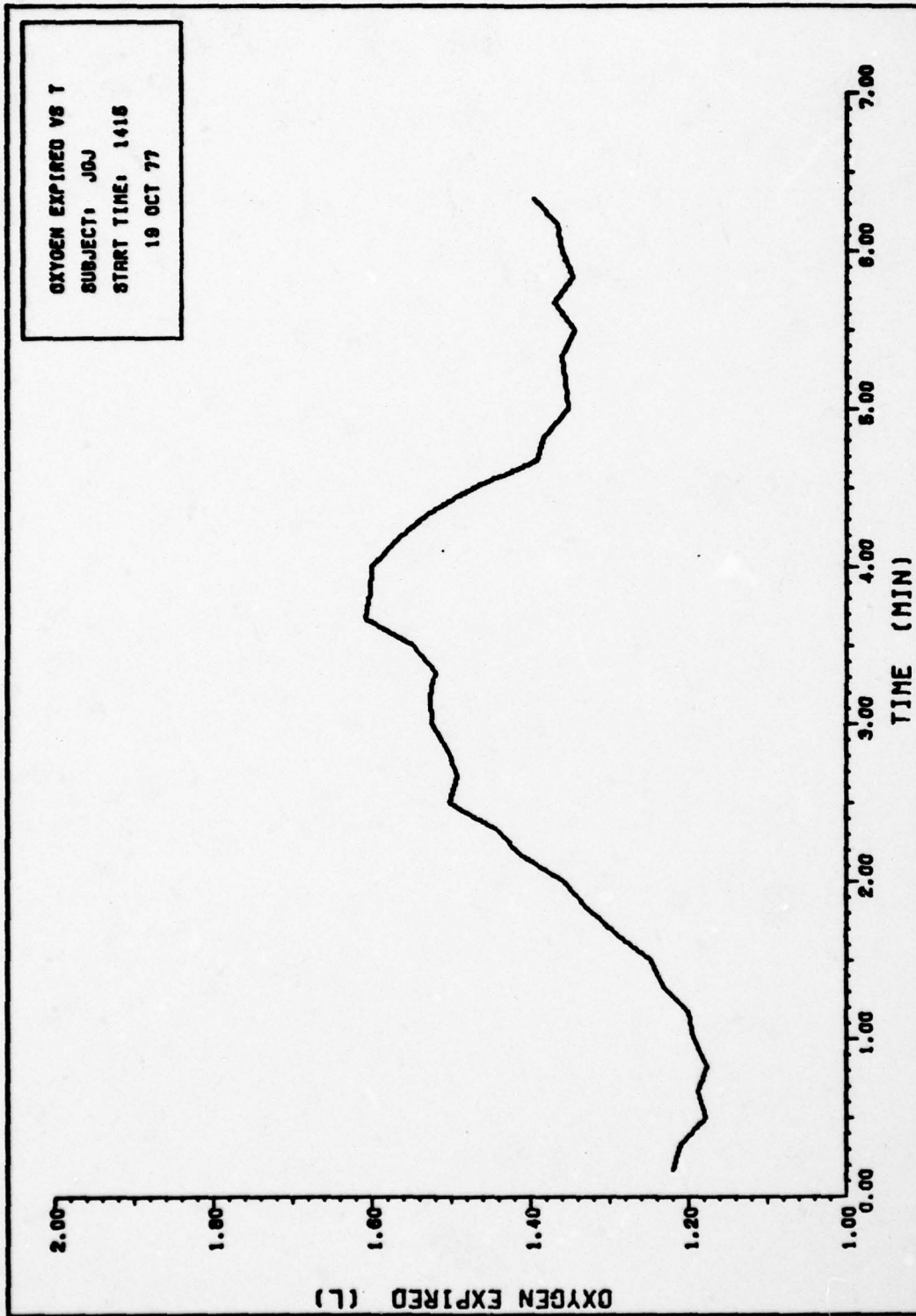


Fig. 14. Expired Oxygen Quantity vs Time

## V System Tests

The IFPDAS II prototype was continually tested during development. Each program module, and each sensor interface, was checked individually both before and after system integration. After all of the modules and available interfaces were integrated, the IFPDAS II prototype was tested as a complete system. The following sections describe the developmental and final tests.

### Developmental Tests

Each of the software modules was designed, coded, and tested on AFIT's main computer system (an 8080 simulator was used to verify the software). The tests consisted of inputting simulated data to check that the modules were manipulating the data properly. After these tests, the programs were transferred to the IFPDAS II prototype's memory.

Then the DAS 1128 was interfaced to the prototype and its conversion and input selection functions were checked. Initially, the  $\overline{\text{STROBE}}$  and  $\overline{\text{TRIG}}$  signals were supplied by digital switches which were used to manually step through the probe selections and start the A/D conversion. Then the operating system-supplied  $\overline{\text{STROBE}}$  and  $\overline{\text{TRIG}}$  signals (through the 8255) were used to perform the same functions. Known DC voltages were used as inputs to the signal multiplexer to check the accuracy of the conversion. For example: with the input voltage at 3.00 volts, the A/D converter output was 96H (150), where:

$$\frac{150}{250} \times 5.00 = 3.00 \quad (17)$$

This testing ensured accurate conversion of the data by the DAS 1128 and proper manipulation of the data by the operating system.

After the operation of the DAS 1128 was verified, the data handling technique of each of the service routines was tested by applying a known DC voltage to the seven analog inputs. For the 3.00 volt example listed above, the routines that sum the readings over the 10 second interval (PO2IN, PO2OUT, and FLWRT) calculated a sum of 7530H (30,000), where,

$$\frac{30,000}{200} = 150 \quad (18)$$

Also, the ABSPR routine stored a reading of 96H. To check the X, Y, and Z-G routines, two different DC voltages (3.00 and 4.00 volts) were used during the 10 second interval. Each of the routines stored 96H as the minimum reading and C8H (200) as the maximum reading. These tests showed that the prototype hardware was functioning as designed and that the operating system and its service routines were exercising the desired control and providing the desired data.

The error handling software was checked by forcing a sensor/service routine mismatch. The operating system identified this error and scheduled the ERROR routine which selected the proper sensor for the service routine. A test was also performed to determine how often the



ERROR routine was being executed. The system was run for four hours which provided over two million chances for a sensor/service routine mismatch. During the test period, the ERROR routine was never executed.

As each sensor interface circuit was developed, it was tested independently. It was then connected to the IFPDAS II prototype to test its integration with the system. Two sensors were not available (absolute pressure and expired flow rate); therefore, their hardware interface and software service routines could not be verified using actual sensor inputs. However, simulated inputs were used for the final tests.

The following sections describe the final tests of the completed IFPDAS II prototype and their results.

#### Final Tests

Heart Rate. The heart rate detector was checked in the following manner. ECG leads were attached to a subject and input to the detector. The output of the ECG amplifier was connected to channel 1 of a Gould Brush 440 strip chart recorder and the output of the R-wave detector was connected to channel 2. This allowed simultaneous recording of the ECG and R-wave detector outputs. Figure 15 shows a strip chart segment; and, a comparison of the IFPDAS II prototype calculated heart rate with a manual computation directly from the chart.

The IFPDAS II prototype uses eight R - R intervals to calculate a representative heart rate. The end of each averaging period was marked on the strip chart using the "mark-event" indicator; and the associated IFPDAS II calculation was recorded.

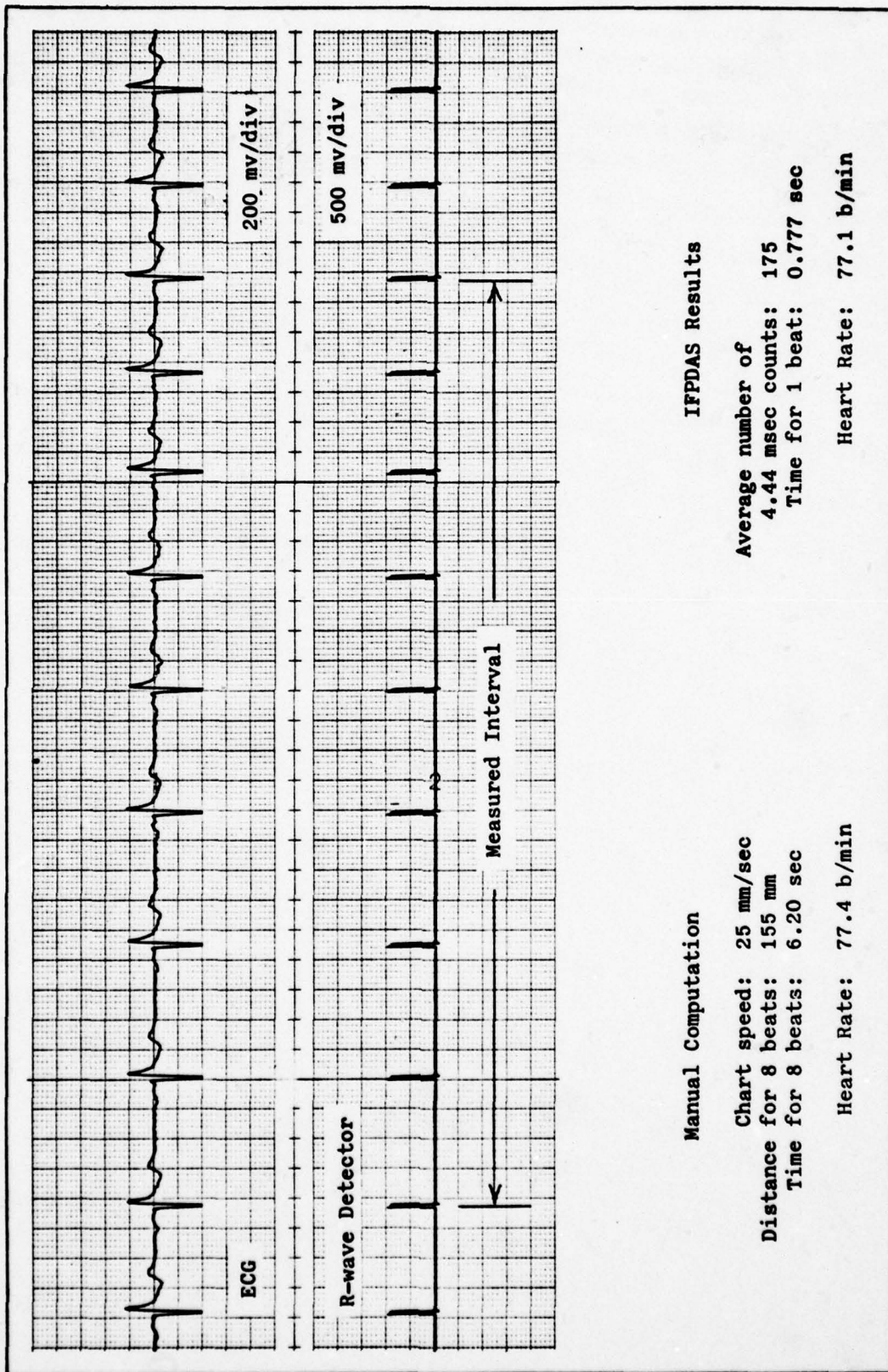


Fig. 15. Heart Rate Computation

The two calculations shown in Figure 15 are within .5% and are typical of the IFPDAS II prototype's accuracy (Appendix A). The differences between the two are due to variances in the chart speed and quantization of the count increment by the prototype.

Oxygen Partial Pressure. The oxygen partial pressure sensor amplifiers were calibrated to indicate 152 mm Hg (1.00 volt). The full scale range was then checked by exposing the OM11 sensor to a pure oxygen source. The amplifier output was 5 volts, indicating 760 mm Hg of oxygen. This 5 volt input (FAH) was properly interpreted and stored by the IFPDAS II operating system.

Acceleration. The Statham F-15-340 accelerometer was interfaced to the IFPDAS II prototype through its amplifier. The sensor axis was oriented to indicate acceleration in the vertical direction and the output of the amplifier was adjusted to 1.33 volts (+1 G). The accelerometer was then rotated 90° to simulate a zero-G situation; and the output dropped to 1 volt (0 G's). Finally, the sensor was rotated an additional 90° to simulate a negative G situation; and the output dropped to .67 volts (-1 G's). The IFPDAS II prototype recorded the minimum and maximum G's (within 0.1 G's) during each ten second interval of the test.

Absolute Pressure. As mentioned earlier, the absolute pressure probe was not available for use on the prototype. For this reason, a known DC voltage was used to simulate the absolute pressure input. As before, the operating system stored the proper value.

Flow Rate. Since the flow rate sensor was not available, a .20 Hz sine wave oscillator was used to supply the flow rate signal. The sine

wave approximates the flow rate signal as shown in Figure 16. (The samples of the negative portion of the sine wave are "0" since the A/D converter input range is 0 - 5 volts.)

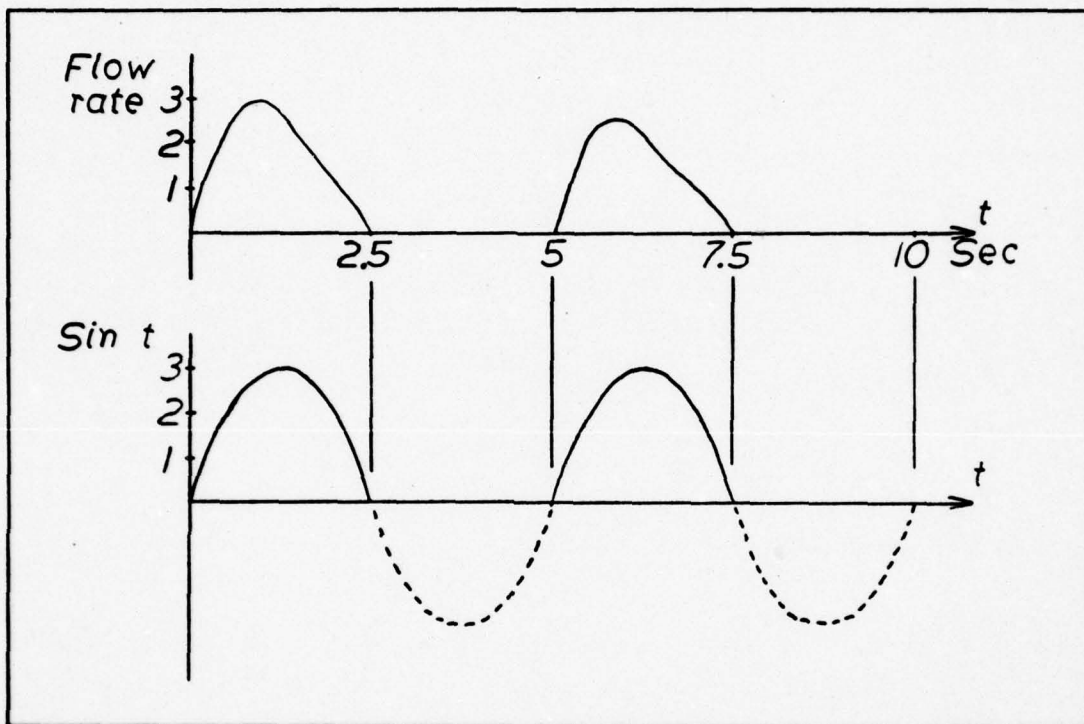


Fig. 16. Sine Wave Approximation of Flow Rate Signal

The amplitude of the sine wave was set to 3.0 volts and the frequency to .20 Hz using an oscilloscope. (The frequency was confirmed by the breathing rate - see the breathing rate test.) The sum of the 200 readings ranged from 2535H to 253EH, averaging 253AH (9530). The corresponding approximation of the integral is.

$$(9530 \text{ counts}) \times .05 \text{ sec} = 476.50 \text{ count-sec} \quad (19)$$

Since there are 50 counts/volt,

$$\frac{476.50 \text{ count-sec}}{50 \text{ counts/volt}} = 9.53 \text{ volt-sec} \quad (20)$$

The exact integral yields:

$$2 \times \int_0^{2.5 \text{ sec}} 3 \sin(2\pi(.2)t) dt = \frac{12}{.4 \pi}$$
$$= 9.55 \text{ volt-sec} \quad (21)$$

The approximation of the integral is within .3% of the actual value. (This technique was also used to check the P02IN and P02OUT summing routines, with the same accuracy.)

Breathing Rate. The flow rate readings are used by the operating system to determine the breathing rate. During the flow rate test, the breathing rate routine recorded 64H (100) fifty msec counts per breath which is one breath every 5 seconds, or 12 breaths per minute.

The test results show that the IFPDAS II prototype monitors the sensor inputs accurately, and can extract and store the desired data. Appendix A summarizes these results and discusses the prototype's accuracy.

## VI Suggestions and Recommendations

The BK10103 Magnetic Bubble Memory System and the TMS9916 controller should be interfaced to the IFPDAS II prototype and tested. This will allow evaluation of the IFPDAS II as a complete independent system.

The data handling technique for the bubble memory system will have to be developed. This includes a software routine to format the data and provide the commands necessary to store the data. In addition, a data retrieval method must be designed. This could include direct transfer from the bubble memory to a large land-based computer; or transfer of the data to a digital cassette tape. Transferring the data to cassette tape might be desirable in the field where access to a large computer is unlikely. The transfer of the entire memory contents would be accomplished by the IFPDAS II's 8080 microprocessor in a matter of minutes, thus, freeing the bubble memory for another mission. The data from the first mission is conveniently stored for later analysis.

The tradeoffs between a magnetic bubble memory system and digital cassette storage systems should be investigated further. This could include acquiring a cassette tape system and interfacing it to the IFPDAS. The performance of both systems could then be directly compared in terms of speed, ease of data transfer, power consumption, etc.

There are several methods to select the "important" data for reduction and storage, in addition to the one selected for the IFPDAS

II prototype. Two others are a breath-by-breath analysis and a "store-on-significant-change" technique. The breath-by-breath analysis would be similar to the method used in the prototype; except that the data would be stored every breath instead of every ten seconds. A digital differentiation technique could be employed to detect significant changes in selected parameters. The operating system would then analyze the change, then collect and store any required data. A thorough analysis of expected flight profiles must be made in order to determine which of the acquisition techniques, or combination of techniques, would be most desirable for a given profile. (This profile analysis may also effect the final choice of memory system.)

An absolute pressure sensor and a bi-directional flow sensor should be acquired and interfaced to the IFPDAS II prototype. With these sensors, and the present OM11 sensors, a more accurate analysis of the oxygen consumption can be made. A suggested method incorporating a single breath analysis is given in Appendix I.

The prototype should be reduced to the IFPDAS II size restrictions. This will include eliminating the unnecessary components on the SBC 80/20 board; and should include consideration of newly developed chips which combine the functions of several chips into one unit. (An example is Intel's new 8085 microprocessor and component family. The components on the SBC 80/20 board that are needed for the IFPDAS II system can be replaced by three 8085 system chips. These components use only 3 watts of power, can be placed on a 3.25" x 4.5" printed circuit board, and are 100% software compatible with the 8080 (Ref 6:109-113).) Once the configuration is reduced, the power consumption should be carefully

considered. At this time, the complete IFPDAS II can be evaluated against the design criteria specified by SAM.

Other areas that should be investigated further include: the capability to record the pilot's voice and synchronize it with the data; a probe check routine which would give a visual indication if the probes were not properly interfaced; and a standard test device to bench-check the entire IFPDAS II operation.



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Appendix A

System Characteristics

**Table I**  
**Parameter Ranges**

Parameter	Desired Range	IFPDAS II Range
Heart Rate	50 to 180 $\pm$ 2 b/min	53 $\pm$ .1 to 225 $\pm$ 2.2 b/min
Breathing Rate	10 to 30 $\pm$ 2 b/min	4.7 $\pm$ .1 to 50 $\pm$ 1 b/min
Flow Rate	N/S	0 to 124 $\pm$ .25 l/min *
Flow Rate Integral	N/S	0 to 26.2 volt-sec ( $\pm$ .3%)
Minute Ventilation Volume	5 to 35 $\pm$ 2 l/min	0 to 100 $\pm$ 2 l/min **
Oxygen Partial Pressure	N/S	0 to 760 $\pm$ 2 mm Hg
Absolute Pressure	N/S	0 to 760 $\pm$ 2 mm Hg *
G's	-3 to +12 $\pm$ .25 G's	-3 to +12 $\pm$ .1 G

N/S = Not Specified

\* Assuming accurate probe input and  $\pm$  1/2 bit accuracy ( $\pm$  10 mv)

\*\* At sea level, assuming an accurate probe input

**Table II**  
**Prototype Power Supplies**

SBC 80/20	+ 5 VDC $\pm$ 5% @ 3.5 Amps
	- 5 VDC $\pm$ 5% @ .180 Amps
	+ 12 VDC $\pm$ 5% @ .467 Amps
	- 12 VDC $\pm$ 5% @ .123 Amps
DAS 1128 and Sensor Interfaces:	+ 15 VDC $\pm$ 3% @ .050 Amps
	- 15 VDC $\pm$ 3% @ .100 Amps
	+ 5 VDC $\pm$ 3% @ .500 Amps

**Table III**  
**Memory Map**

Hexadecimal Address	Use
0000 - 03DF	Operating System
03E0 - 03FF	Interrupt Jump Table
0400 - 0FFF	Unused PROM
1000 - 37FF	Not Used
3800 - 3BFF	Unused RAM
3C00 - 3C2F	CPU Scratch Pad Area
3C30 - 3F4F	Unused RAM
3F50 - 3F80	CPU Stack Area
3F81 - 3FFF	Unused RAM

Table IV

Pin Connections for DAS 1128 Interface

SBC 80/20 Board		Component Board	
Signal	J1 Pin Connection	J4 Pin Connection	Signal
Port 2 - Bit 7	2	52	8 Out
Bit 6	4	54	4 Out
Bit 5	6	56	2 Out
Bit 4	8	58	1 Out
Bit 3	10	60	B9
Bit 2	12	62	B10
Bit 1	14	64	B11
Bit 0	16	66	B12 (LSB)
IBFA	18	68	N/C
<u>STROBE</u>	20	70	<u>STROBE</u>
IBFB	22	72	N/C
<u>TRIG</u>	24	74	<u>TRIG</u>
<u>STBA</u>	26	76	<u>EOC</u>
N/C	28	78	N/C
N/C	30	80	N/C
<u>STBB</u>	32	82	<u>EOC</u>
Port 1 - Bit 7	34	84	B1 (MSB)
Bit 6	36	86	B2
Bit 5	38	88	B3
Bit 4	40	90	B4
Bit 3	42	92	B5
Bit 2	44	94	B6
Bit 1	46	96	B7
Bit 0	48	98	B8
N/C	50	100	N/C

Odd numbered pins are GND

Table V

Pin Connections for Heart Rate Detector Interface

SBC 80/20 Board		Component Board	
Signal	J2 Pin Connection	J4 Pin Connection	Signal
Port 2 - Bit 7	2	2	N/C
Bit 6	4	4	N/C
Bit 5	6	6	N/C
Bit 4	8	8	N/C
Bit 3	10	10	N/C
Bit 2	12	12	N/C
Bit 1	14	14	N/C
Bit 0	16	16	N/C
IBFA	18	18	N/C
N/C	20	20	N/C
IBFB	22	22	N/C
N/C	24	24	N/C
$\overline{\text{STBA}}$	26	26	$\overline{\text{STB}}$
N/C	28	28	N/C
N/C	30	30	N/C
$\overline{\text{STBB}}$	32	32	N/C
Port 1 - Bit 7	34	34	B1 (MSB)
Bit 6	36	36	B2
Bit 5	38	38	B3
Bit 4	40	40	B4
Bit 3	42	42	B5
Bit 2	44	44	B6
Bit 1	46	46	B7
Bit 0	48	48	B8 (LSB)
225 Hz Clock	50	50	Clock

Odd numbered pins are GND



Table VI

Hazeltine Interface Connections

Signal	J3 Pin Connection	Hazeltine Connection	Signal
Protective GND	2	1	Protective GND
Transmit Data	4	2	Transmit Data
Receive Data	6	3	Receive Data
Request to Send	8		N/C
Clear to Send	10		N/C
Signal GND	14	7	Signal GND
+12 volts	22	5	Clear to Send
		6	Data Set Ready
		8	Data Carrier Detect

Unused pins are not shown

Table VII

<u>Other Characteristics</u>	
Maximum analog inputs	: 16
Analog input impedance	: $> 10^{10}$ ohms
Analog input voltage	: 0 - 5.12 volts
Sampling rate	: 20 Hz
Machine cycle time	: 465 nsec
Timer clock period	: 930 nsec
Heart rate clock	: 225 Hz
Transfer (baud) rate	: 1200 baud

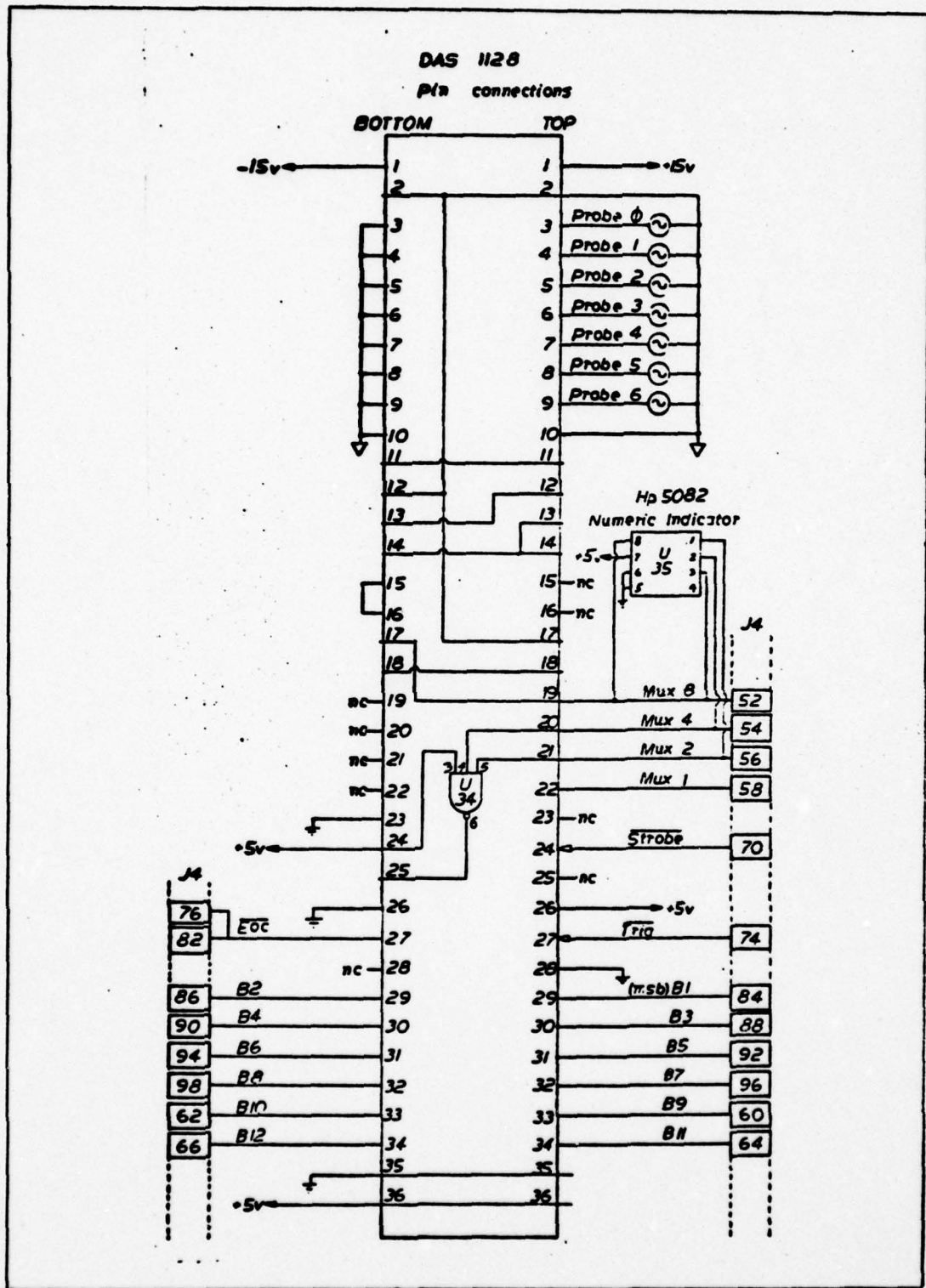


Fig. 17. DAS 1128 Interconnections and Interface

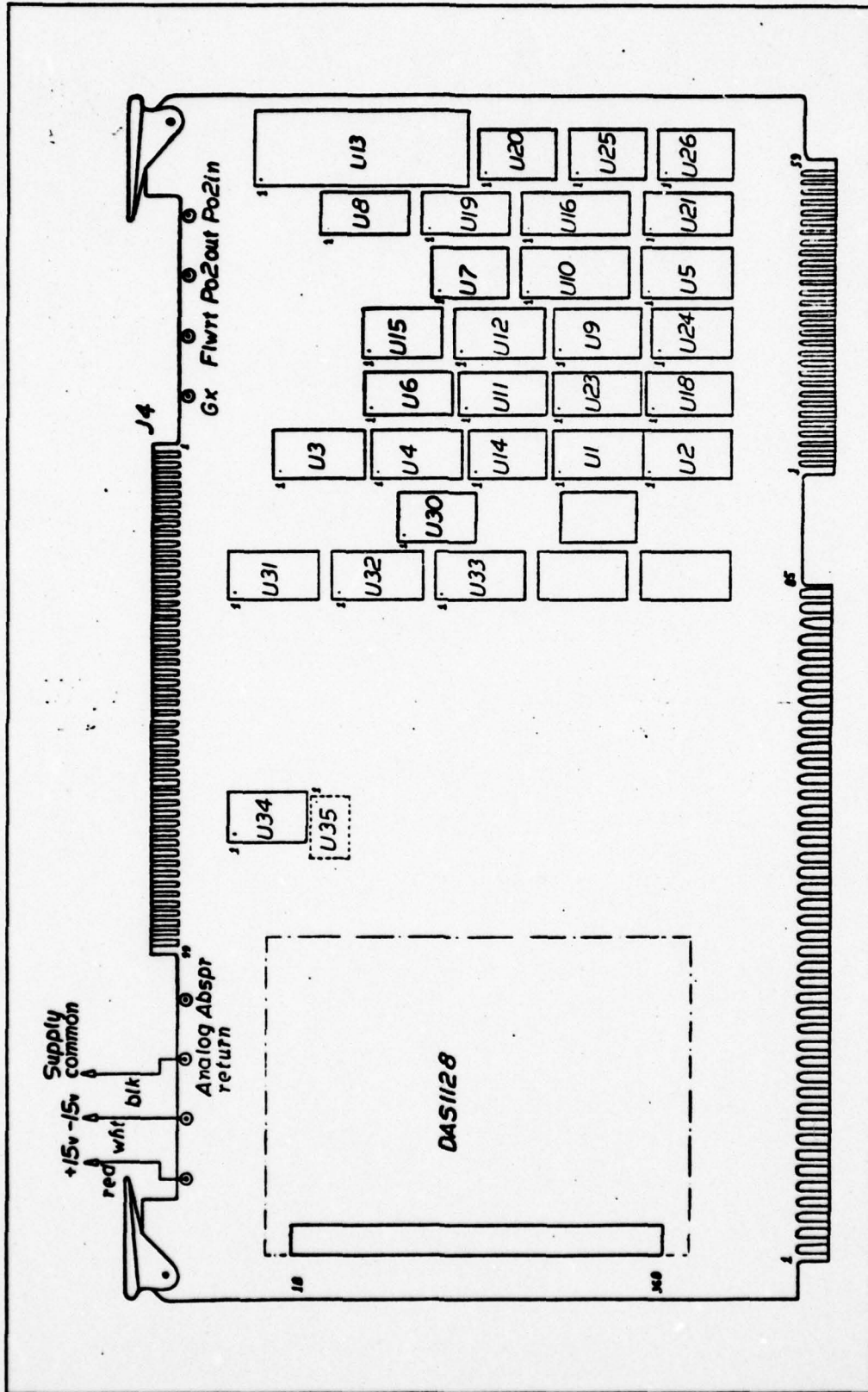


Fig. 18. Component Board Layout

## Appendix B

### Operating System

The operating system controls the operation of the IFPDAS II prototype hardware. Execution of the operating system by the CPU results in initialization of the system and acquisition, reduction, and storage of the data. The operating system selects the proper input sensor, starts the A/D conversion, and inputs the data for reduction and permanent storage. In addition, it ensures that the proper service routine receives the data.

A flow chart of the operating system starts on this page; a listing of the program follows the flow chart.

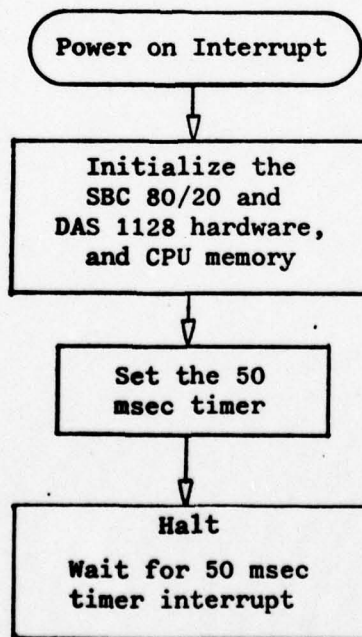


Fig. 19. Operating System Flow Chart (Sheet 1 of 9)

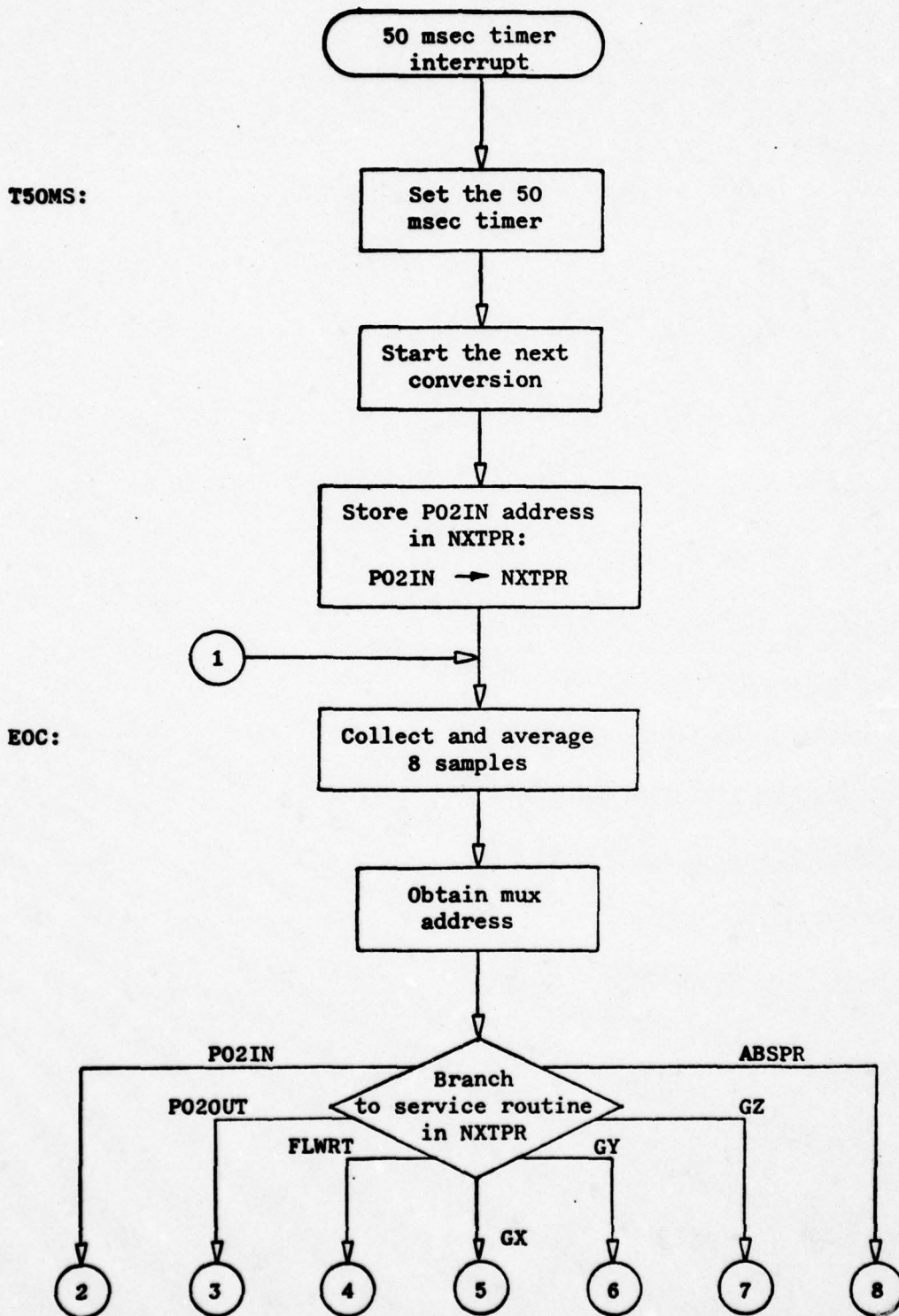
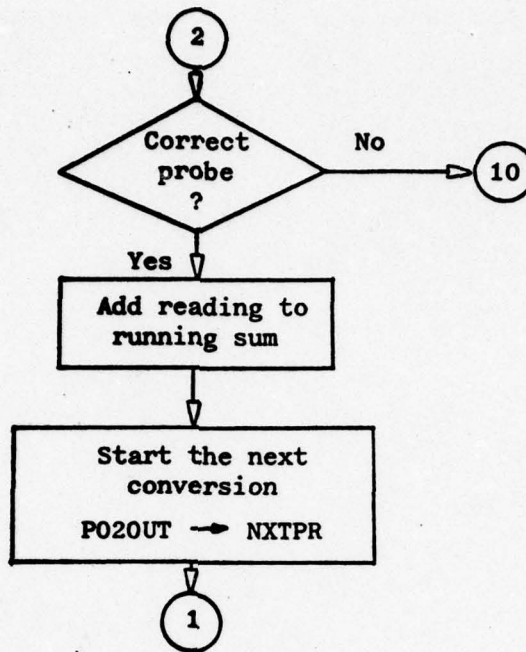


Fig. 19. Operating System Flow Chart (Sheet 2 of 9)

PO2IN:



PO2OUT:

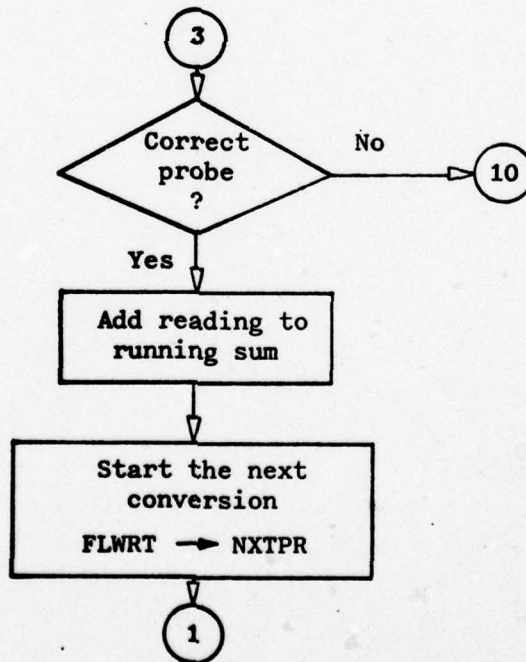
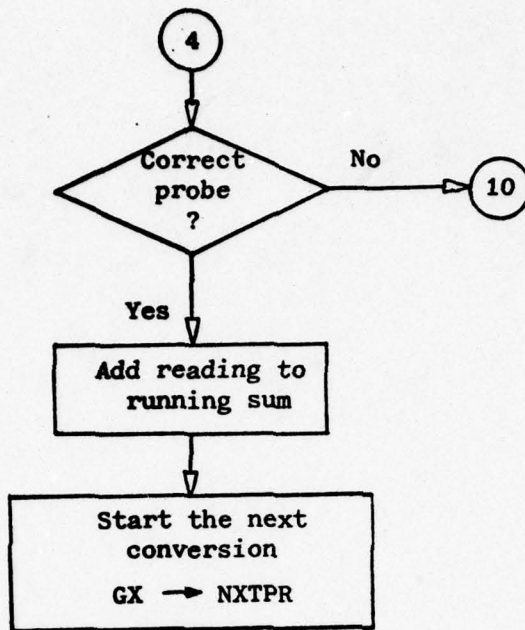


Fig. 19. Operating System Flow Chart (Sheet 3 of 9)

FLWRT:



BRRT:

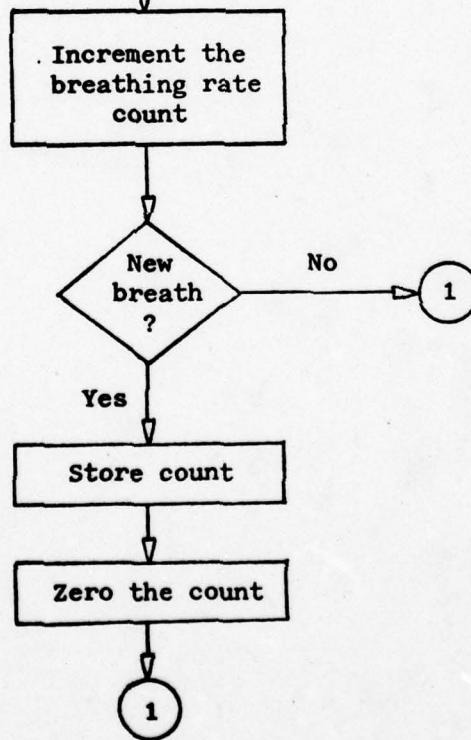


Fig. 19. Operating System Flow Chart (Sheet 4 of 9)

GX:

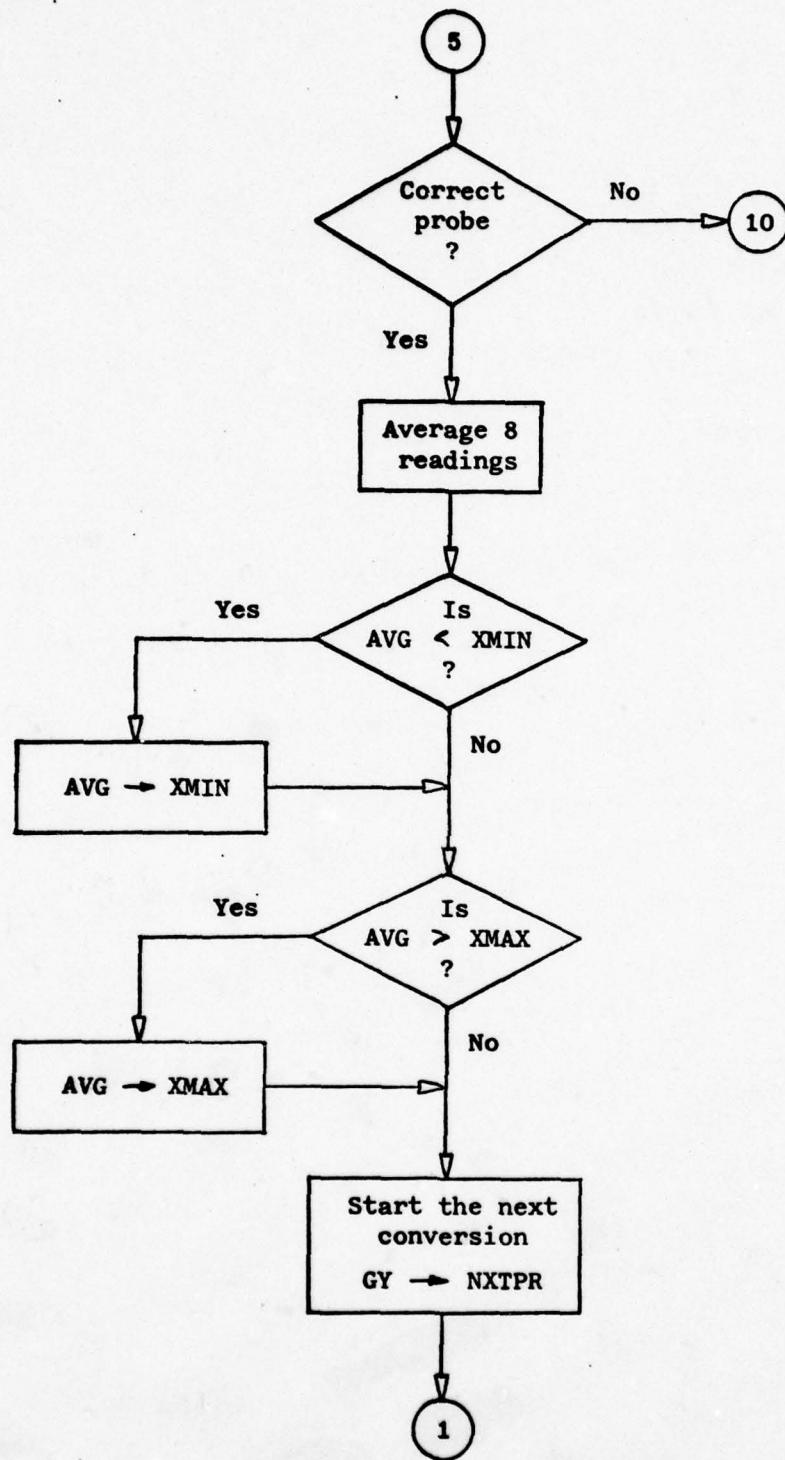


Fig. 19. Operating System Flow Chart (Sheet 5 of 9)



GY:

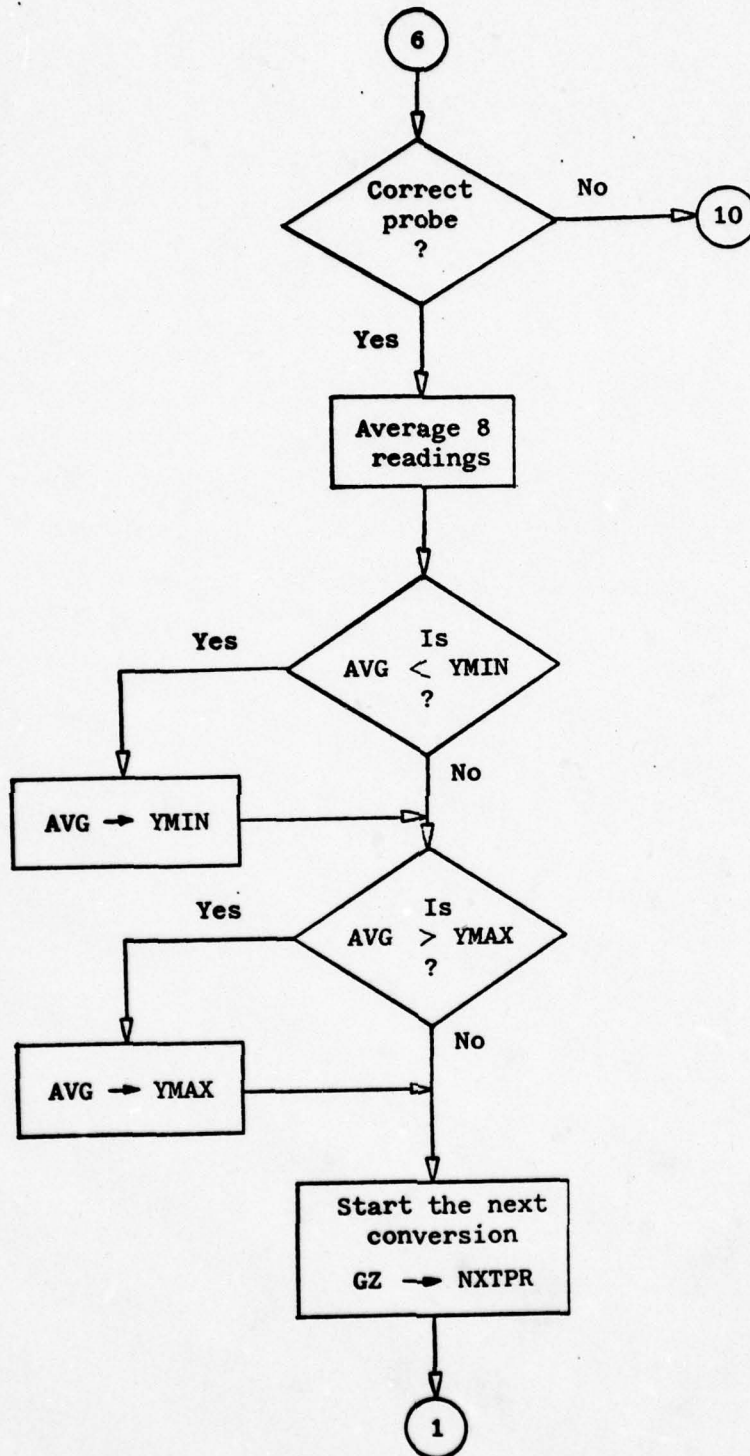


Fig. 19. Operating System Flow Chart (Sheet 6 of 9)

GZ:

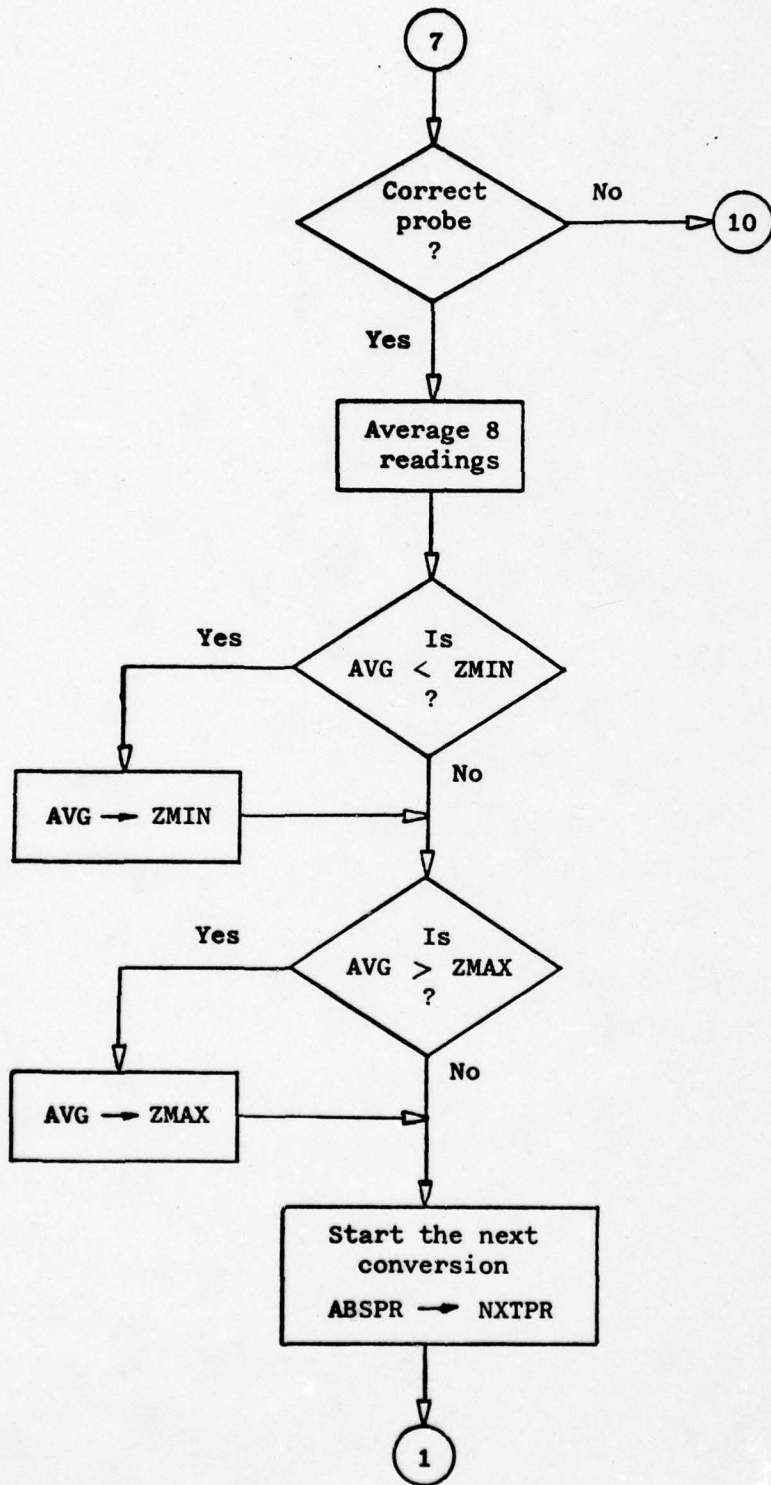
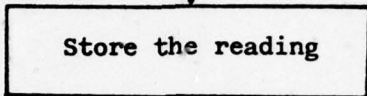
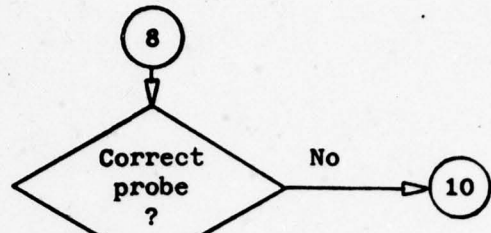
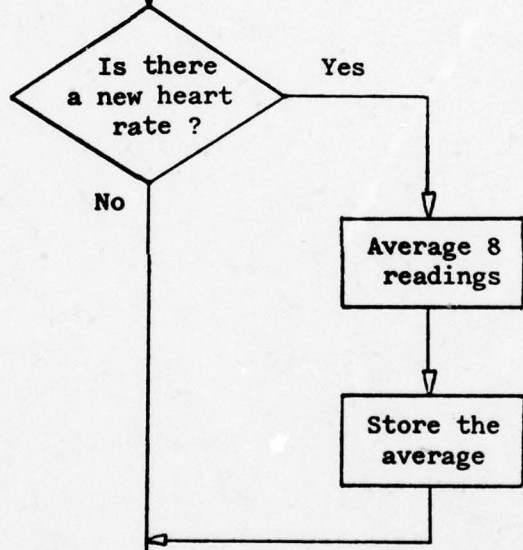


Fig. 19. Operating System Flow Chart (Sheet 7 of 9)

ABSPR:



HEART:



CNTCK:

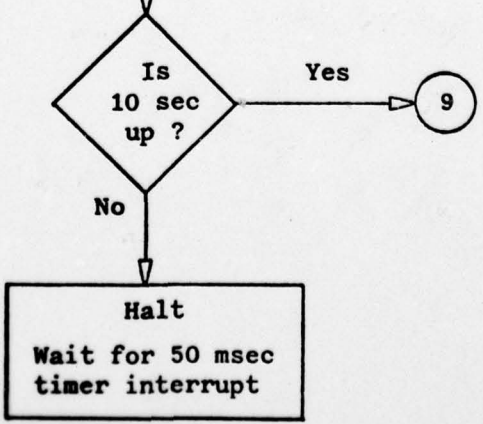
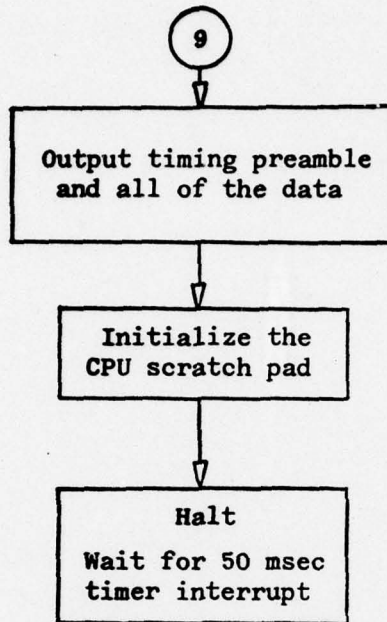


Fig. 19. Operating System Flow Chart (Sheet 8 of 9)

STORE:



ERROR:

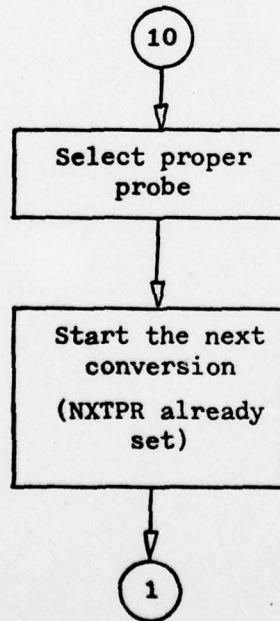


Fig. 19. Operating System Flow Chart (Sheet 9 of 9)

IFPDAS II PROTOTYPE  
OPERATING SYSTEM

THIS PROGRAM CONTROLS THE OPERATION OF THE IFPDAS II PROTOTYPE HARDWARE. EXECUTION OF THE OPERATING SYSTEM BY THE CPU RESULTS IN INITIALIZATION OF THE SYSTEM AND ACQUISITION, REDUCTION, AND STORAGE OF THE DATA. THE OPERATING SYSTEM SELECTS THE PROPER INPUT SENSOR, STARTS THE A/D CONVERSION, AND INPUTS THE DATA FOR REDUCTION AND PERMANENT STORAGE. IN ADDITION, IT ENSURES THAT THE PROPER SERVICE ROUTINE RECEIVES THE DATA.

THE OPERATING SYSTEM CONSISTS OF THE FOLLOWING PROGRAM MODULES:

PWRUP - CONFIGURES THE IFPDAS II HARDWARE FOR THE DATA ACQUISITION FUNCTION

T50MS - RESETS THE 50 MSEC TIMER AND STARTS THE PROGRAM LOOP

EOC - OBTAINS THE DATA AND PASSES IT TO THE APPROPRIATE SERVICE ROUTINE

PO2IN - INSPIRED OXYGEN PARTIAL PRESSURE SERVICE ROUTINE

PO2OUT - EXPIRED OXYGEN PARTIAL PRESSURE SERVICE ROUTINE

FLWRT - FLOW RATE SERVICE ROUTINE

BRRT - BREATHING RATE CALCULATIONS

GX - X-G'S SERVICE ROUTINE

GY - Y-G'S SERVICE ROUTINE

GZ - Z-G'S SERVICE ROUTINE

ABSPR - ABSOLUTE PRESSURE  
SERVICE ROUTINE

HEART - HEART RATE CALCULATIONS

CNTCK - TEN SECOND COUNTER

STORE - OUTPUTS THE DATA EVERY  
10 SECONDS

ERROR - CORRECTS SERVICE ROUTINE  
PROBE MISMATCHES

THE OPERATING SYSTEM INCLUDES THE  
FOLLOWING SUBROUTINES:

STRTAD - CALLS STROBE AND TRIGGR

STROBE - INCREMENTS THE DAS 1128  
MUX ADDR

TRIGGR - STARTS THE A/D CONVERSION

SUM - PERFORMS DOUBLE PRECISION  
ADDITION

AVG - PERFORMS DATA AVERAGING

INIT - INITIALIZES CPU SCRATCH PAD

DTOUT - PREPARES DATA FOR OUTPUT AND  
CALLS ASCII AND PRINT

ASCII - CONVERTS BINARY DATA TO ASCII

PRINT - OUTPUTS ASCII CHARACTER  
TO THE CONSOLE

ASSEMBLER EQU STATEMENTS

```

:
:
:
:
:
:
:
:
:
:
:
004E      MODE      EQU      04EH      :MODE SFT FOR USART INIT
0037      RSTURT   EQU      037H      :CMD INST TO RESET USART
00E0      USART    EQU      0E0H      :USART CONTROL PORT
:
:
0001      READY    EQU      01H       :MASK FOR TRANSMITTER READY
00EC      COM      EQU      0ECH      :CONSOLE OUTPUT PORT
:
:
3F80      STACK    EQU      3F80H     :INITIAL STACK POINTER VALU
:
:
00F4      DATA    EQU      0E4H      :MUX ADDR INPUT PORT
:          :          :          :   (PORT A OF 8255 #1)
00E5      MUX      EQU      0E5H      :CONVERTED DATA INPUT PORT
:          :          :          :   (PORT B OF 8255 #1)
00E6      STAT1    EQU      0E6H      :STATUS PORT OF 8255 #1
00E7      PPI1     EQU      0E7H      :8255 #1 CONTROL PORT
00E3      HRIN     EQU      0E6H      :HEART RATE INPUT PORT
:          :          :          :   (PORT A OF 8255 #2)
00EA      STAT2    EQU      0EAH      :STATUS PORT OF 8255 #2
00EA      PPI2     EQU      0EBH      :8255 #2 CONTROL PORT
0096      AI7I     EQU      0B6H      :8255 CONTROL WORD:
:          :          :          :   A - MODE 1, INPUT
:          :          :          :   B - MODE 1, INPUT
0094      AI90     EQU      094H      :8255 CONTROL WORD:
:          :          :          :   A - MODE 1, INPUT
:          :          :          :   B - MODE 1, OUTPUT
:
010C      C5OFF    EQU      0CH       :TURNS OFF C6
010E      C7OFF    EQU      0EH       :TURNS OFF C7
0123      A7FSK    EQU      020H      :MASK TO OBTAIN IBFA
0122      B9FSK    EQU      0B2H      :MASK TO OBTAIN OBF8
:
:
012D      ST9ON    EQU      004H      :TURNS ON ST90E (SETS C6)
010C      ST9OFF   EQU      0CH       :TURNS OFF ST90E (C6 OFF)
010F      TR9ON    EQU      0FH       :TURNS ON TRIGGER (SETS C7)
010E      TR9OFF   EQU      0EH       :TURNS OFF TRIGGER (C7 OFF)
01E7      A0CON    EQU      0E7H      :PSEUDO ADDR FOR DAS 1128
:          :          :          :   VIA PORT C OF 8255 #1
:
:
0036      C043     EQU      036H      :CTR 0 TO MODE 3
0070      C14C     EQU      070H      :CTR 1 TO MODE 0
0076      C243     EQU      0B6H      :CTR 2 TO MODE 3
010C      CTR0     EQU      00CH      :COUNTER 0 PORT
0100      CTR1     EQU      0004      :COUNTER 1 PORT
010E      CTR2     EQU      00E4      :COUNTER 2 PORT
010F      TMCP     EQU      00FH      :TIMER COMMAND PORT

```

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00F6	ICW1	EQU	0F6H	: INTERRUPT CMD WORD 1
0007	ICW2	EQU	003H	: INTERRUPT CMD WORD 2
000A	ICCP1	EQU	00AH	: INT CONTROLLER CMD PORT 1
000E	ICCP2	EQU	00BH	: INT CONTROLLER CMD PORT 2
0004	IMASK	EQU	04H	: INT MASK VALUE
				: MASKS OFF LEVEL 2
0008	MSKPT	EQU	008H	: INT MASK PORT
	:			
0020	TRHLD	EQU	20H	: BREATHING RATE THRESHOLD
00FF	FLAG	EQU	0FFH	: SET FLAG VALUE
	:			
00F0	MUXMSK	EQU	0F0H	: MASK FOR MUX ADDR
0060	PROBNT	EQU	60H	: PROBE COUNT (#0 - 6)
0020	ENDIC	EQU	020H	: END OF INT CMD WRD
0011	BYTES	EQU	017D	: # OF OUTPUT BYTES
0009	LOCS	EQU	11D	: # OF LOCS TO BE ZEROED
	:			
007F	CSP	EQU	7EH	: ASCII CODE FOR 'CS.'
001E	PR	EQU	1EH	: ASCII CODE FOR 'PRINT'



POWER UP ROUTINE

THE POWER-UP ROUTINE IS EXECUTED WHEN  
THE HARDWARE RECOGNIZES THAT POWER HAS  
BEEN APPLIED TO THE IFPDAS. (THE POWER-  
UP RESET TRIGGERS OFF THE +5 VOLT SUPPLY.)  
THE 8251 COMMUNICATION INTERFACE (USART),  
8255 I/O PORTS, 8253 TIMERS, 8259  
INTERRUPT CONTROLLER, DAS 1128 DATA  
ACQUISITION MODULE, AND THE CPU SCRATCH  
PAD STORAGE AREA ARE ALL INITIALIZED.

THE POWER-UP ROUTINE STARTS THE  
50 MSEC TIMER: WHEN IT TIMES OUT, THE  
PROGRAM LOOP IS ENTERED AND THE DATA  
ACQUISITION FUNCTION IS BEGUN.



8255 INITIALIZATION

```

8255 #1
MODE 1 (STROBED I/O) 86 :
SET MODE
PORT A - MODE 1
PORT A - INPUT
C6,C7 - OUTPUT
PORT B - MODE 1
PORT B - INPUT
BITS C6 & C7 OFF
    
```

```

8255 #2
MODE 1 (STROBED I/O) 94 :
SET MODE
PORT A - MODE 1
PORT A - INPUT
C6,C7 - OUTPUT
PORT B - MODE 1
PORT B - OUTPUT
    
```

```

0009 3E95
0000 03E7
000F 3E0C
0011 03E7
0013 3E0E
0015 03E7
0017 3E94
0019 03E9
    
```

```

MVI A,A181 :OUTPUT CONTROL WORD
OUT PPI1 : TO 8255 #1
MVI A,C6OFF :ENSURE BIT C6 IS OFF
OUT PPI1
MVI A,C7OFF :ENSURE BIT C7 IS OFF
OUT PPI1
MVI A,A180 :OUTPUT CONTROL WORD
OUT PPI2 : TO 8255 #2
    
```

INTERRUPT INITIALIZATION

```

ICM1 F6 : JUMP TABLE AT
ICM2 03 : 03E3 TO 03FF
INTERVAL = 4
IMASK 04 : MASK OFF LEVEL 2
    
```

```

0019 3EF5
0017 0304
001E 3E03
0021 0307
0023 3E04
002F 0303
    
```

```

MVI A,ICM1
OUT ICCP1 :OUTPUT COMMAND WORD 1
MVI A,ICM2
OUT ICCP2 :OUTPUT COMMAND WORD 2
MVI A,IMASK
OUT MSKPT :OUTPUT MASK WORD
    
```





```

: *****
: *
: * THIS SECTION INITIALIZES THE
: * STORAGE AREA. TIME, PINSM
: * POUTS, HRTRT, GXMAX, SYMAX, &
: * GZMAY ARE ZEROED. THEN OFFH
: * IS LOADED INTO GXMIN, SYMIN,
: * & GZMIN. 2000 IS LOADED
: * INTO THE TEN SEC COUNTER.
: *
: *****
:
:
:

```

```

0059 AF          IDA:  XRA      A
005C 210J3C     LXI      H,TIME   :ZERO TIME COJNT BYTE 1
005F 77         MOV      M,A
0060 23         INX      H        :ZERO TIME COUNT BYTE 2
0061 77         MOV      M,A
0062 C0E702     CALL     INIT    :INITIALIZE PINS4 TO GZMIN
0065 21133C     LXI      H,TENSC
006A 36C9       MVI      M,200D   :SETS THE COUNTER TO 200
006A AF         XRA      A
0069 160F       MVI      D,15
006D 23         INA1:  INX      H        :ZEROES THE EVENT COUNTERS
006E 77         MOV      M,A        : AND TEMPORARY RUNNING
006F 15         DCR      D        : SUMS
0070 C25000     JNZ      IDA1

```

```

: *****
: *
: * THIS SECTION STARTS THE 50 MS
: * TIMER. WHEN THE TIMER EXPIRES
: * THE PROGRAM LOOP IS ENTERED
: * AT T50MS.
: *
: *****
:
:
:

```

```

0073 3EES       MVI      A,0E5H
0075 0300     OUT     CTR1    :MOVES E5 TO LSB OF COUNTER
0077 3E01     MVI      A,0D1H
0079 0300     OUT     CTR1    :MOVES D1 TO MSB OF COUNTER
0079 F9       EI
007C 76       HLT
007D 00       NOP

```

PROGRAM LOOP

THE PROGRAM LOOP IS EXECUTED WHEN THE 50 MSEC TIMER EXPIRES. IT CONTAINS THE SERVICE ROUTINES - ONE FOR EACH PHYSIOLOGICAL AND ENVIRONMENTAL SENSOR. THE DATA IS COLLECTED BY THE END-OF-CONVERSION (EOC) ROUTINE WHICH AVERAGES 6 SAMPLES INTO 1 READING WHICH IT PASSES TO THE APPROPRIATE SUBROUTINE.

THE "END-OF-CONVERSION" IS DETECTED BY CHECKING THE "INPUT-BUFFER-FULL" (IRF) STATUS BIT. THIS METHOD WAS USED, INSTEAD OF HAVING THE 8255 INTERRUPT THE CPU, TO ELIMINATE THE OVERHEAD TO HAVE THE CPU HANDLE THE INTERRUPT. THE TIME TO COMPLETE THE CONVERSION IS KNOWN, 25 MICRO-SEC: AND, NORMALLY, THE TIME FROM THE TRIGGER TO THE CHECK OF THE IRF BIT IS MORE THAN 25 MICRO-SEC.





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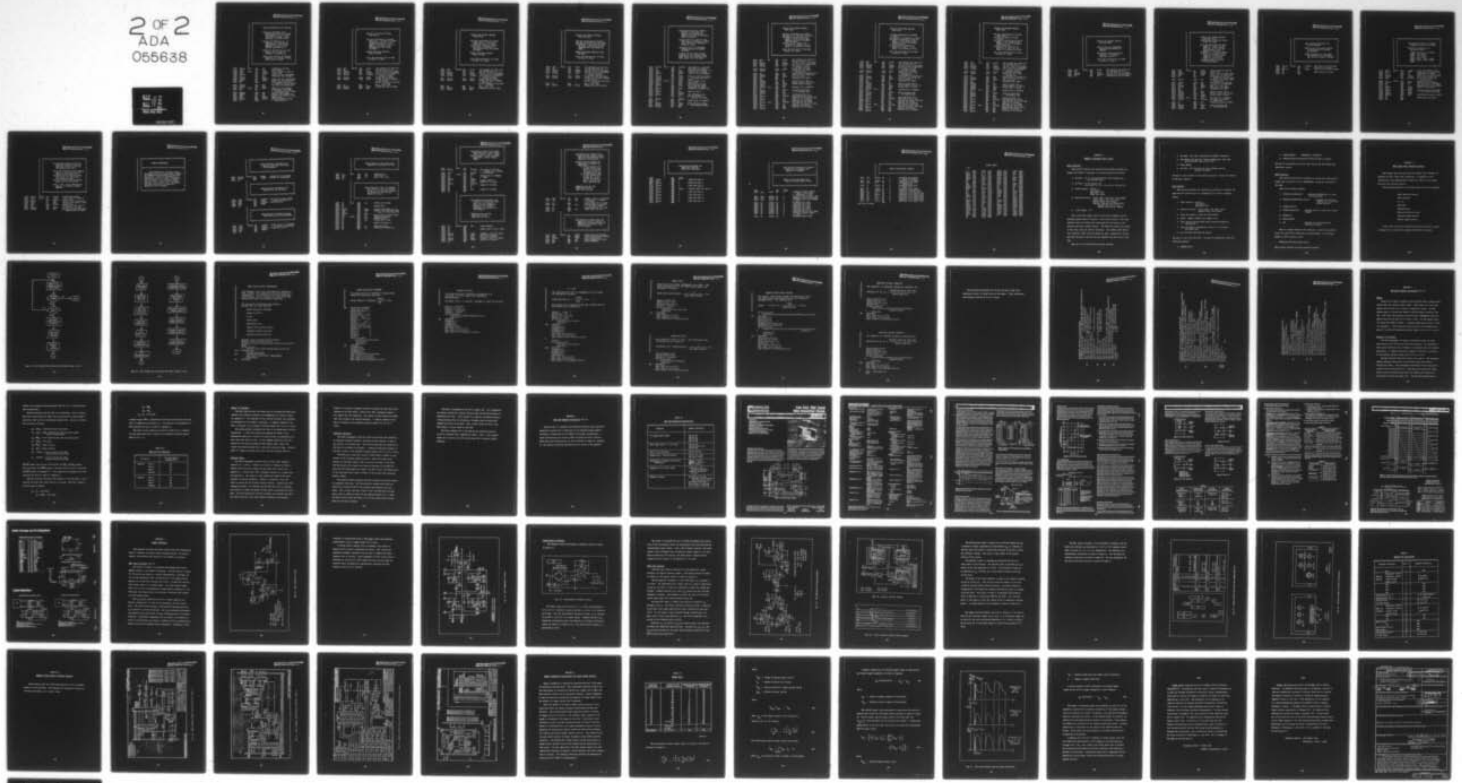
AIR FORCE INST OF TECH WRIGHT-PATTERSON AFB OHIO SCH--ETC F/G 6/16  
AIRCREW INFLIGHT PHYSIOLOGICAL DATA ACQUISITION SYSTEM II.(U)  
DEC 77 J G JOLDA, S J WANZEK

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AFIT/GE/EE-77-21

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THIS IS THE SERVICE ROUTINE  
FOR X-G'S

THE NEXT CONVERSION IS STARTED  
THE ADDR OF GY IS STORED IN NXTPR  
THE CURRENT READING OF X-G'S  
INPT. EIGHT READINGS ARE  
AVERAGED TO GIVE AN AVERAGE  
X-G'S READING.

IF THE AVG .GT. GXMAX IT IS  
IS STORED IN GXMAX - IF .LT.  
GXMIN IT IS STORED IN GXMIN

THE A REG CONTAINS THE MUX ADDR  
IN BITS 7-4 (XOH)

0140	0630	GX1	MVI	9,3CH	:PUT DESIRED MUX ADDR IN 9
014F	98		CMP	9	:IS THE MUX ADDR 9 ?
0150	C29302		JNZ	ERROR	:IF NOT, THERE IS AN ERROR
0153	C79302		CALL	STRAD	:START THE NEXT CONVERSION
0156	219501		LXI	M,GY	:GET ADDR OF "Y-G'S"
					: SERVICE ROUTINE
0159	22113C		SHLD	NXTPR	:STORE ADDR IN NXTPR
015C	2A143C		LHLC	XG	:LOADS HSL WITH X-G'S
					: RUNNING SUM
015F	78		MOV	A,E	:MOVES CURRENT READING TO A
0160	C0C302		CALL	SUM	:ADD DATA TO RUNNING SUM
0163	341C3C		LOA	XGCNT	:GET X-G COUNT
0166	3C		INR	A	:INCREMENT THE COUNT
0167	FE03		CPI	8	
0169	C47301		JZ	GX1	:JUMP IF XGCNT .EQ. 8
016C	22143C		SHLD	XG	:STORE RUNNING SUM IN XG
016F	321C3C		STA	XGCNT	:STORE COUNT IN XGCNT
0172	C39C00		JMP	EOC	
0175	C7CE02	GX1:	CALL	AVG	:AVERAGE THE 8 READINGS
0179	AF		XRA	A	
0179	321C3C		STA	XGCNT	:ZERO THE COUNT AND
017C	32143C		STA	XG	: THE RUNNING SUM
017F	32173C		STA	XG+1	
0182	78		MOV	A,E	:MOVES AVG DATA TO A
0183	21093C		LXI	M,GXMAX	:GET ADDR OF MAX X-G'S
0186	9E		CMP	M	:COMPARE AVG VALUE WITH MAX
0187	0A9301		JC	GX2	:JUMP IF A .LT. GXMAX
018A	77		MOV	M,A	:REPLACE IF A.GE.GXMAX
0189	21093C	GX2:	LXI	M,GXMIN	:GET ADDR OF MIN X-G'S
018F	9E		CMP	M	:COMPARE AVG VALUE WITH MIN
0190	029C00		JNC	EOC	:JUMP IF A.GE.GXMIN
0192	77		MOV	M,A	:REPLACE IF A.LT.GXMIN
0193	C39C00		JMP	EOC	









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```

: *****
: *
: *
: * THIS IS THE SERVICE ROUTINE
: * FOR HEART RATE
: * THIS ROUTINE IS ENTERED RIGHT
: * AFTER A3S PRESS
: *
: * IF A NEW HEART RATE HAS BEEN
: * COMPUTED (I9FB=1) IT IS
: * READ IN. EIGHT READINGS
: * ARE AVERAGED TO GIVE AN
: * AVERAGE HEART RATE.
: * READINGS .LT. 10H ARE ASSUMED
: * AS FALSE R-WAVES AND ARE
: * ADDED TO THE COUNT TO BE
: * AVERAGED - BUT DO NOT
: * COUNT AS ONE OF THE 8.
: * THE NEW AVERAGE HEART RATE
: * IS THEN STORED.
: *
: *
: *****

```

```

0232 09EA HEART: IN STAT2 :INPUT STATUS OF 8255 #2
0234 E620 ANI ABFMSK :MASKS IBFA
0235 CA6A02 JZ CNTCK :JUMP IF NO NEW HEART RATE
0239 03E9 IN HRIN :INPUT THE NEW HEART RATE
0239 2F CMA :COMPLEMENTS DATA FROM
: INVERTING DRIVER (8226)
023C 2A143C LHL0 HRT :LOADS HSL WITH HEART RATE
: RUNNING SUM
023F 00C902 CALL SUM :ADD DATA TO RUNNING SUM
0242 FE10 CPI 10H : (A HAS THE NEW READING)
0244 0A6A02 JC CNTCK :DONE IF A .LT. 10H
0247 3A153C LDA HRCNT :GET HEART RATE COUNT
024A 3C INR A :INCREMENT THE COUNT
0249 FE09 CPI 8
024D CA5902 JZ HRT1 :JUMP IF HRCNT .EQ. 8
0250 22143C SHLD HRT :STORE RUNNING SUM IN HRT
0253 32153C STA HRCNT :STORE COUNT IN HRCNT
0256 C35A02 JMP CNTCK
0259 00CE02 HRT1: CALL AVG :AVERAGE THE 8 READINGS
025C 210F3C LXI H,HRTRT :GET ADDR OF CURRENT
: HEART RATE
025F 73 MOV M,E :STORE NEW VALUE IN HRTRT
0260 AF XRA A
0261 32153C STA HRCNT :ZERO THE COUNT AND
0264 32143C STA HRT : THE RUNNING SUM
0267 32153C STA HRT+1

```

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```
*****  
*  
* THIS SECTION DECREASES THE *  
* 10 SEC COUNTER *  
*  
* EACH 10 SEC THE STORAGE ROUTINE *  
* IS CALLED AND THE COUNTER *  
* IS RELOADED *  
*  
* AFTER COMPLETION OF THIS CODE, *  
* THE PROGRAM HALTS AND WAITS *  
* FOR THE 50 MS TIMER TO EXPIRE *  
*  
*****
```

```
026A 21133C CNTCK: LXI 4,TENSC :GET ADDR OF 10 SEC COUNT  
0267 35 DCR M :DECREMENT THE COUNT  
026F CA7:02 JZ STORE :JUMP IF 10 SEC HAS ELAPSED  
0271 F2 EI  
0272 76 HLT :WAIT FOR 50 MS TIMER  
0273 00 NCP
```





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SYSTEM SUBROUTINES

THESE SUBROUTINES ARE USED BY THE  
OPERATING SYSTEM AND THE SERVICE ROUTINES  
TO ACCOMPLISH TASKS COMMON TO MORE THAN  
ONE MODULE. THERE ARE SUBROUTINES TO  
CONTROL THE DAS 1128 (START A CONVERSION  
AND INCREMENT THE MUX ADDRESS), MANIP-  
ULATE THE DATA (SUMMING AND AVERAGEING)  
INITIALIZE THE STORAGE AREA, AND OUTPUT  
THE DATA (FORMATTING THE DATA AND  
SENDING IT TO THE CONSOLE

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```
*****
*
* THIS SUBROUTINE INCREMENTS THE
* DAS 1123 MUX ADDR AND STARTS
* THE CONVERSION
*
*****
```

```
0290 C09702 STRTAD: CALL STROBE ;INCREMENT THE MUX ADDR
0293 C0C002 CALL TRIGGR ;STARTS THE CONVERSION
0296 C9 RET
```

```
*****
*
* THIS SUBROUTINE INCREMENTS THE
* MUX ADDR OF THE DAS 1128
*
*****
```

```
0297 3E03 STROBE: MVI A,STBON ;SETS BIT C6 WHICH INCRE-
0299 03E7 OUT ADCON ; MENTS MUX ADDR BY 1
029A 3E0C MVI A,STROFF ;RESETS BIT C5
029D 03E7 OUT ADCON
029F C9 RET
```

```
*****
*
* THIS SUBROUTINE TRIGGERS THE DAS
* 1128 WHICH STARTS A CONVERSION
*
*****
```

```
02C0 3E0F TRIGGR: MVI A,TRGON ;TURNS ON BIT C7 (TRIGGER)
02C2 03E7 OUT ADCON ; TO START A CONVERSION
02C4 3E0E MVI A,TRGOFF ;TURNS OFF BIT C7
02C5 03E7 OUT ADCON
02C8 C9 RET
```











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```

: .....
: *
: *
: * THIS SECTION CONTAINS THE STORAGE *
: * AREA FOR THE PROGRAM (CPU *
: * SCRATCH PAD STORAGE) *
: *
: * .....
: *
: * THESE ARE THE LOCATIONS THAT *
: * WILL BE STORED EVERY 10 SEC *
: *
: * .....
:

```

3C10			ORG	3C90H	
3C3C	0000	TIME:	0W	0	:CONTAINS RUNNING OUTPUT : COUNT. EACH COUNT : REPRESENTS 10 SEC
3C02	0000	PINS4:	0W	0	:CONTAINS P02IN RUNNING SUM
3C04	0000	POUTS:	0W	0	:CONTAINS P02OJT RUNNING SM
3C06	0000	FLRTS:	0W	0	:CONTAINS FLOW RATE RUNNING : SUM
3C08	00	GX4AX:	0B	0	:CONTAINS MAX X-G'S
3C0A	00	GX4IN:	0B	0	:CONTAINS MIN X-G'S
3C0C	00	GY4AX:	0B	0	:CONTAINS MAX Y-G'S
3C0E	00	GY4IN:	0B	0	:CONTAINS MIN Y-G'S
3C10	00	GZ4AX:	0B	0	:CONTAINS MAX Z-G'S
3C12	00	GZ4IN:	0B	0	:CONTAINS MIN Z-G'S
3C14	00	CA3PR:	0B	0	:CONTAINS CURRENT ABS PRESS
3C16	00	HRTRT:	0B	0	:CONTAINS CURRENT HEART : RATE
3C18	00	BTHRT:	0B	0	:CONTAINS BREATH RATE



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SYMBOL TABLE

\* 01

A	3007	AFMS	0020	A9SPR	0228	A9CON	00E7
AI3I	3095	AI3C	00F4	ASCII	0311	AVG	02CE
AVG1	3200	B	0000	BFMS	0002	BRRT	3111
BRRT1	3125	BTHPT	3C10	BTES	0C11	C	3001
C043	3035	C140	0070	C2M3	0036	C6OFF	000C
C70FF	300E	CA3PR	3C0E	CNTCK	025A	CON	00EC
CSP	307E	CTR0	000C	CTR1	000D	CTR2	300E
D	3002	DATA	00E4	DTOUT	02FE	E	3003
EOC	309C	FO21	00A1	EOC2	0038	EOIC	0020
ERR1	329E	ERROR	0299	FLAG	00FF	FLRTS	3C06
FLWPT	30F8	GX	0140	GX1	0175	GX2	0189
GXMAX	3C09	GXMIN	3C09	GY	0196	GY1	018E
GY2	3104	GYMAX	3C0A	GYMIN	3C09	GZ	010F
G71	3207	G72	0210	GZMAX	3C0C	GZMIN	3C0D
H	3004	HEART	0232	HRCNT	3C16	HRIN	30E8
HRT	3C14	HRT1	0259	HRTRT	3C0F	ICCP1	30DA
IC002	3003	ICW1	00F6	ICW2	0003	IDA	0059
IDA1	3060	I4ASK	00C4	INIT	02E7	INIT1	02E0
L	3005	LOGS	000B	M	0006	MODE	004E
MSKPT	3009	MUX	00E5	MUXMS	00F0	NXTPR	3C11
PINS4	3C02	PO2IN	00C9	PO2OU	00E2	POUTS	3C04
PPI1	30E7	POI2	00F8	PR	001E	PRRCV	3060
PRINT	3319	PS4	0006	PWRUP	004E	READY	3001
RSTHD	3037	SP	0006	STACK	3F30	STAT1	00E6
STAT2	30EA	STR0F	000C	STR0N	000D	STOR1	0282
STOPF	3274	STR0R	0287	STRTA	0290	SUM	32C9
T534S	307E	TENSC	3C13	TFLAG	3C19	THCNT	3C18
TIME	3000	THCNT	3C17	THCP	007F	TRGOF	030E
TRGON	300F	TRHLD	0020	TRIGG	020C	USART	30ED
XG	3C1A	XGCNT	3C1C	YG	3C10	YGCNT	3C1F
ZG	3C20	ZGCNT	3C22				

## Appendix C

### IFPDAS II Prototype User's Guide

#### Data Collection

This section describes the step-by-step procedure necessary to operate the IFPDAS II prototype in its data acquisition function.

1. AC Power - ON to both the Hazeltine video terminal and the cassette tape
2. DC Power - +5 VDC supply: OFF  
all others (-5, +12, -12, +15, & -15 VDC): ON
3. Video Terminal - Parity: 1  
Full Duplex  
Baud Rate: 1200  
Clear the screen
4. Cassette Recorder - Insert tape in Recorder 2 and engage  
Select CONT - OFF LINE - PAGE  
Depress RESET, then REWIND  
Depress INTERLOCK and RECORD button  
and wait for tape to stop  
(RECORD button stays lighted)
5. +5 VDC Supply - ON

The +5 VDC power supply resets the SBC 80/20 hardware and the operating system starts to execute. Data will be output to the video terminal every ten seconds and transferred from the screen to the cassette tape under program control. The 8080 CPU directs the writing of this data using two control characters. The CONTROL SHIFT PERIOD (cs.) character (7EH) tells the Hazeltine that a command will follow. The PRINT character (1EH) directs the transfer from the screen to the tape.

When all of the desired data has been recorded:



1. DC Power - OFF (this terminates the IFPDAS's operation)
2. Push RESET (the tape will advance momentarily, then stop and the RECORD light will go out)
3. Press REWIND
4. AC Power - OFF (to both the video terminal and the cassette recorder)

The data is now recorded on the cassette tape and is ready for transfer to the main computer.

#### Data Transfer

This section describes the step-by-step procedure to transfer the data from the cassette tape to a permanent file on AFIT's computer system.

1. Video Terminal - Parity: 1  
Half Duplex  
Baud Rate: 300
2. Cassette Recorder - Select CONT - OFF LINE - PAGE  
Depress RESET, then REWIND
3. Using the terminal, LOGIN and enter EDITOR
4. Enter: CREATE, SUPPRESS line numbers (C,S)
5. After system responds ENTER LINES, depress PLAYBACK on tape channel
6. After the data is transferred, send an "=" to release the CREATE mode
7. List the file and check for errors

The data is now in the edit file. To store it permanently, enter the following commands:

1. REQUEST,Q,\*PF

2. SAVE,Q,NOSEQ,0 (NOSEQuence, Overwrite)
3. CATALOG,Q,DATA,ID=(problem #),RP=(# of days to retain)

The data is now stored on disc for later use by the post-flight data conversion routine.

### Data Conversion

This section describes the procedure to execute the compiled post-flight data conversion routine (COMP\_CONVERT), using the file DATA as the data.

Enter the following commands:

1. ATTACH,LGO,COMP\_CONVERT (attaches COMP\_CONVERT as a local file called LGO)
2. ATTACH,AFITSUBROUTINES,ID=AFIT (attaches the AFIT sub-routines as a local file called AFITSUB)
3. LIBRARY,AFITSUB
4. ATTACH,TAPE10,DATA (attaches DATA as a local file called TAPE10)
5. REWIND,LGO
6. REWIND,TAPE10
7. LGO (executes the post-flight data conversion routine)

When the program completes its execution, it will have created a local file called PLOT containing the output graphs. To send these graphs to AFIT's plotter, enter:

```
ROUTE,PLOT,TID=BB,FID=(xxx),DC=PT
```

This routing completes the data conversion process.

## Appendix D

### Post Flight Data Conversion Routine

This program takes the data from the cassette tape (TAPE10) and performs the post flight data conversions. In addition to the calculations, each program module formats the labels for the graphs and calls the plotting routines.

The following parameters are plotted versus time by this program:

Cabin Absolute Pressure

Cabin Altitude

Z - G's

Heart Rate

Breathing Rate

Minute Ventilation Volume

Inspired Oxygen Quantity

Expired Oxygen Quantity

A flow chart of the post flight data conversion routine is given in Figure 20; a listing of the program follows the flow chart.

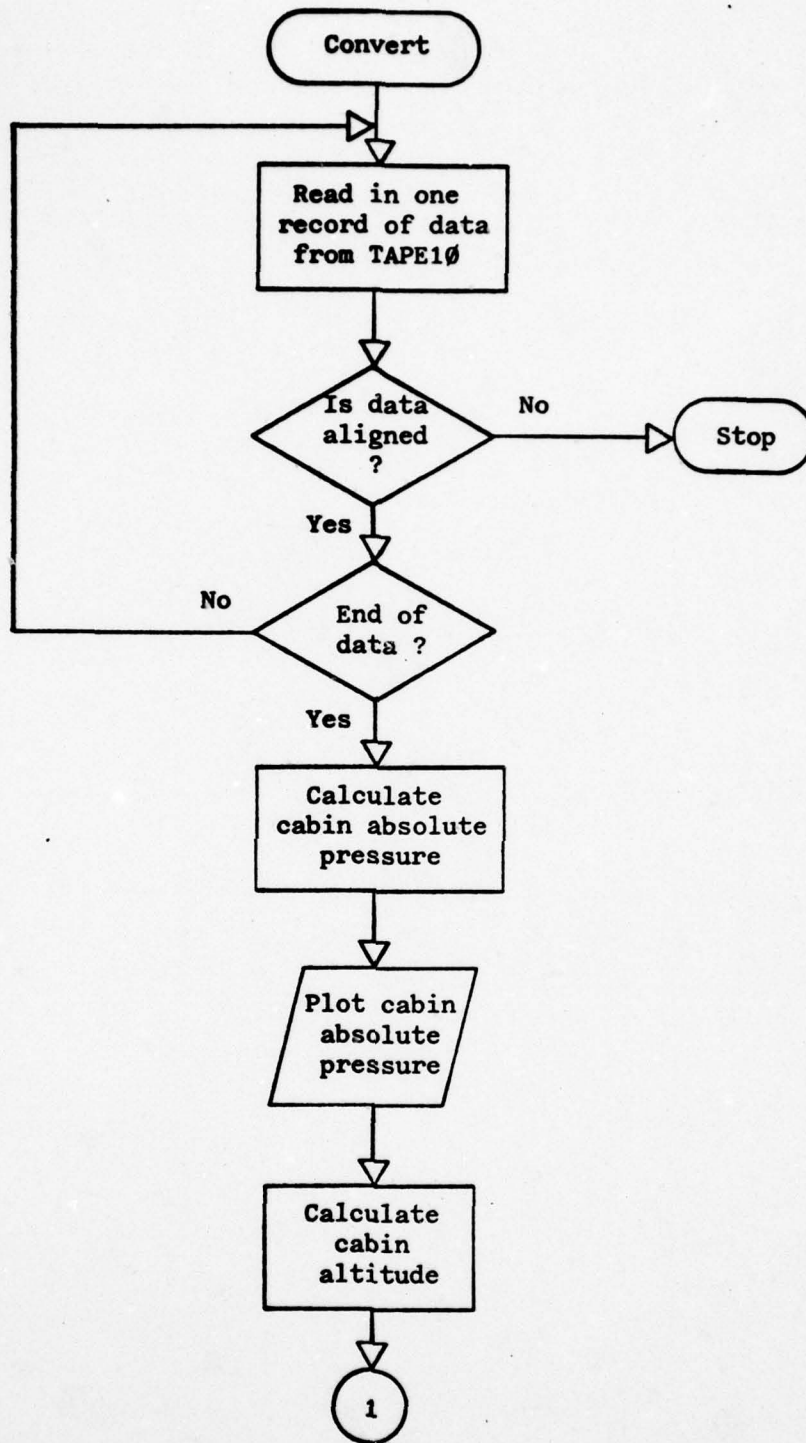


Fig. 20. Post Flight Data Conversion Flow Chart (Sheet 1 of 2)

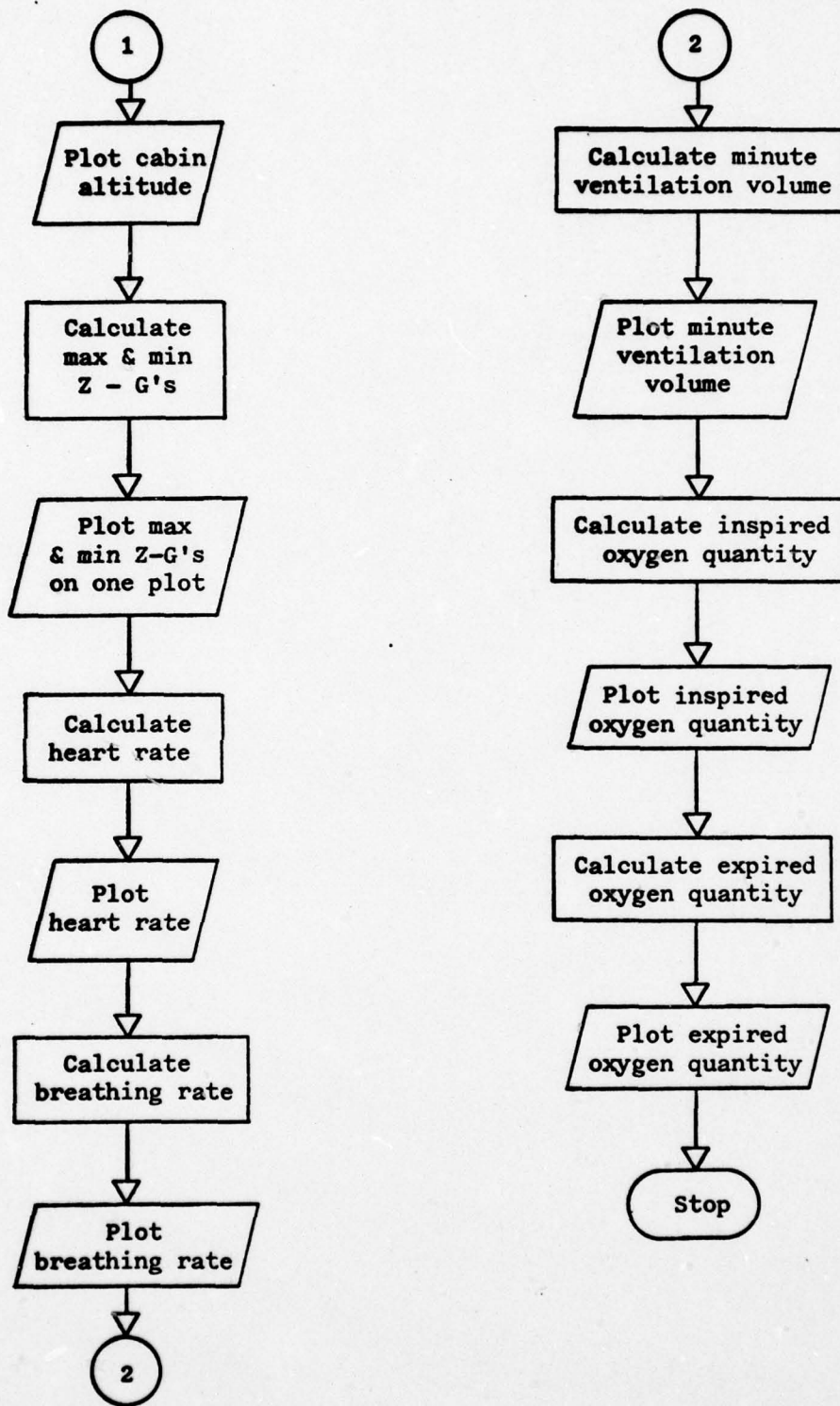


Fig. 20. Post Flight Data Conversion Flow Chart (Sheet 2 of 2)





C  
C  
C  
C  
C  
C  
C  
C

CABIN ALTITUDE

THE CABIN ABSOLUTE PRESSURE IS CONVERTED TO  
ALTITUDE (IN FEET) USING THE FOLLOWING  
RELATIONSHIP:

$$\text{ALTITUDE (FEET)} = 170,156 - 25,685 \text{ LN (ABS PR (MM HG))}$$

ID(1)=" ALTITUDE"

ID(2)=" VS TIME "

ID(11)=" ALTITUDE"

ID(12)=" (FEET) "

DO 65 I=1,NROW

Y(I)=170516-25685\*(ALOG(3(I,1)))

X(I)=I/6.

65

CONTINUE

X(NROW+1)=0.

Y(NROW+1)=0.

X(NROW+2)=1.

Y(NROW+2)=5000.

CALL PLOT (0.,-4.,-3)

CALL PLOT (0.,0.03,-3)

CALL HGRAPH (X,Y,NROW,ID,-1,0,0)



C  
C  
C  
C  
C  
C  
C  
C  
C  
C  
C  
C

Z - G'S

THE ACCELERATION DATA IS CONVERTED TO G'S USING  
THE FOLLOWING FORMULA:

$$\text{ACCELERATION(G'S)} = \left( \frac{\text{(DATA)}}{250} \times 15 \right) - 3$$

THE MAXIMUM AND MINIMUM VALUES ARE PLOTTED ON THE  
SAME GRAPH FOR COMPARISON.

```

ID(1)="  Z-G'S  "
ID(2)="VS  TIME  "
ID(11)="MAX AND MI"
ID(12)="N  Z-G'S  "
DO 120 I=1,NROW
      Y(I)=(A(I,11)+.06)-3.
      X(I)=I/6.
120  CONTINUE
      X(NROW+1)=0.
      Y(NROW+1)=0.5
      X(NROW+2)=1.
      Y(NROW+2)=.4
      CALL PLOT (0.,-4.,-3)
      CALL PLOT (0.,0.03,-3)
      CALL HGRAPH (X,Y,NROW,ID,-1,0,0)
C
ID(1)=0
DO 130 I=1,NROW
      Y(I)=(A(I,12)+.06)-3.
      X(I)=I/6.
130  CONTINUE
      X(NROW+1)=0.
      Y(NROW+1)=0.5
      X(NROW+2)=1.
      Y(NROW+2)=.4
      CALL PLOT (0.,-4.,-3)
      CALL PLOT (0.,0.03,-3)
      CALL HGRAPH (X,Y,NROW,ID,2,1,1)
```

C  
C  
C  
C  
C  
C  
C  
C  
C  
C

### HEART RATE

EACH HEART RATE COUNT REPRESENTS 4.44 MSEC. THE  
COUNT IS CONVERTED TO THE HEART RATE USING THE  
FOLLOWING EQUATION:

$$\text{HEART RATE (BEATS/MIN)} = \frac{1}{(4.44 \text{ MSEC}) (\text{COUNT})} \times 60$$

ID(1)=" HEART RAT"  
ID(2)="E VS TIME "  
ID(11)="HEART RATE"  
ID(12)=" (B/MIN) "  
DO 40 I=1,NROW  
    Y(I)=(225.\*60.)/A(I,14)  
    X(I)=I/6.  
40 CONTINUE  
CALL PLOT(0.,-4.,-3)  
CALL PLOT(0.,0.03,-3)  
CALL HGRAPH(X,Y,NROW,ID,1,0,0)

C  
C  
C  
C  
C  
C  
C  
C  
C

### BREATHING RATE

EACH COUNT IS WORTH 50 MSEC. THE BREATHING RATE  
IS CALCULATED AS FOLLOWS:

$$\text{BREATHING RATE (BREATHS/MIN)} = \frac{1}{(50 \text{ MSEC}) (\text{COUNT})} \times 60$$

ID(1)="BREATH RAT"  
ID(11)="BREATH RAT"  
ID(12)="E (B/MIN) "  
DO 50 I=1,NROW  
    Y(I)=(20.\*60.)/A(I,15)  
    X(I)=I/6.  
50 CONTINUE  
CALL PLOT(0.,-4.,-3)  
CALL PLOT(0.,0.03,-3)  
CALL HGRAPH(X,Y,NROW,ID,1,0,0)

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C  
C  
C  
C  
C  
C  
C  
C  
C  
C

### MINUTE VENTILATION VOLUME

THE MINUTE VENTILATION VOLUME IS CALCULATED USING  
THE IPPOAS-SUPPLIED SUMMATION AND THE ABSOLUTE  
PRESSURE DATA. THE FORMULA IS:

$$\text{VOLUME 10 SEC (L)} = \frac{(\text{SUM})}{\text{SQRT(ABS PR)}} \times .0114$$

```
DO 70 I=1,NROW
    B(I+3,2)=(A(I,5)+256+A(I,6))/(SQRT(B(I,1))+87.75)
70  CONTINUE
    B(1,2)=B(4,2)
    B(2,2)=B(4,2)
    B(3,2)=B(4,2)
    B(NROW+4,2)=B(NROW+3,2)
    B(NROW+5,2)=B(NROW+3,2)
DO 80 I=1,NROW
    B(I,2)=B(I,2)+B(I+1,2)+B(I+2,2)+B(I+3,2)+B(I+4,2)
    +B(I+5,2)
    Y(I)=B(I,2)
    X(I)=I/6.
80  CONTINUE
    ID(1)="MIN VENT V"
    ID(2)="OL VS TIME"
    ID(11)="MIN VENT V"
    ID(12)="OL (L)  "
    CALL PLOT (0.,-4.,-3)
    CALL PLOT (0.,0.03,-3)
    CALL HGRAPH (X,Y,NROW,ID,1,0,0)
```

INSPIRED OXYGEN QUANTITY

THE QUANTITY OF INSPIRED OXYGEN IS COMPUTED BY:

$$\text{QUANTITY OF O}_2 \text{ (L)} = \frac{(\text{PC}_2\text{IN SUM})(\text{MIN VENT VOL})}{65.79 \text{ (ABS PR)}}$$

C  
C  
C  
C  
C  
C  
C

```
DO 90 I=1,NROW
  ID(1)="OXYGEN INT"
  ID(2)="AKE VS T "
  ID(11)="OXYGEN INT"
  ID(12)="AKE (L) "
  Y(I)=((A(I,1)+256*A(I,2))*3.04*B(I,2))
  / (200.*B(I,1))
```

90

```
  X(I)=I/6.
CONTINUE
CALL PLOT (0.,-4.,-3)
CALL PLOT (0.,0.03,-3)
CALL HGRAPH (X,Y,NROW,ID,1,0,0)
```

EXPIRED OXYGEN QUANTITY

THE QUANTITY OF EXPIRED OXYGEN IS COMPUTED BY:

$$\text{QUANTITY OF O}_2 \text{ (L)} = \frac{(\text{PC}_2\text{OUT SUM})(\text{MIN VENT VOL})}{65.79 \text{ (ABS PR)}}$$

C  
C  
C  
C  
C  
C  
C

```
DO 110 I=1,NROW
  ID(1)="OXYGEN EXP"
  ID(2)="IRED VS T "
  ID(11)="OXYGEN EXP"
  ID(12)="IRED (L) "
  Y(I)=((A(I,3)+256*A(I,4))*3.04*B(I,2))
  / (200.*B(I,1))
```

110

```
  X(I)=I/6.
CONTINUE
CALL PLOT (0.,-4.,-3)
CALL PLOT (0.,0.03,-3)
CALL HGRAPH (X,Y,NROW,ID,1,0,0)
```

C

```
STOP "YOU MADE IT"
END
```

The following subroutines are used by the post flight data conversion routine to format and plot the graphs. These subroutines were borrowed from AFIT's EE 6.91 course.

```
10 SUBROUTINE HGRAPH(X,Y,N,ID,NO,NP,NS)
   DIMENSION X(1),Y(1),ID(1) $ IF (NO.EQ.2) CALL PLOT(-1.05,2.10,-3)
   IF (NO.EQ.2) GO TO 30 $ IF (NO.LT.0) GO TO 10
   CALL SCALE(X,7.,N,1) $ CALL SCALE(Y,5.,N,1)
   CALL PLOT(0.,11.,2) $ CALL PLOT(8.5,11.,2)
   CALL PLOT(8.5,0.,2) $ CALL PLOT(0.,0.,2)
   CALL PLOT(1.35,1.35,-3) $ CALL PLOT(0.,8.30,-2)
   IF(ID(1).EQ.000) GO TO 25
   CALL PLOT(.1,-.1,-3) $ CALL PLOT(0.,-2.,-2)
   DO 20 I=1,7,2
   CALL SYM30L( (I+1.5)*.1,.3,.07,ID(I),90.,20)
   CALL PLOT(0.,0.,2) $ CALL PLOT(1.,0.,2)
   CALL PLOT(1.,2.,2) $ CALL PLOT(0.,2.,-2)
   CALL PLOT(-.1,.1,-3)
   CALL PLOT(5.8,0.,-2)
   CALL PLOT(0.,-8.35,-2) $ CALL PLOT(-5.8,0.,-2)
   CALL SYM30L(.5,-.2,1,1,(13),0.,50) $ CALL PLOT(5.3,.75,-3)
   CALL AXIS(0.,0.,ID(9),-20,7.,90.,X(N+1),X(N+2))
   CALL AXIS(0.,0.,ID(11),20,5.,180.,Y(N+1),Y(N+2))
   Y(N+2)=-Y(N+2) $ CALL LINE(Y,X,N,1,NP,NS)
   Y(N+2)=-Y(N+2) $ CALL PLOT(1.85,-2.10,-3)
   RETURN
```

```
10 SUBROUTINE AXIS(X0,Y0,L,NC,RL,ANG,RMIN,DR)
   DIMENSION L(1) $ A=ANG*3.14159/180. $ DX=.1*COS(A) $ DY=.1*SIN(A)
   IC=ISIGN(1,NC) $ NNC=IARS(NC) $ R=.1 $ N=1 $ X=X0 $ Y=Y0 $
   CALL PLOT(X,Y,3) $ X=X+DX $ Y=Y+DY $ CALL PLOT(X,Y,2)
   CALL PLOT(X-.21+DY+IC,Y+.21+DX+IC,2)
   IF(N.EQ.5) CALL PLOT(X-.42+DY+IC,Y+.42+DX+IC,2)
   IF(N.EQ.10) CALL PLOT(X-.70+DY+IC,Y+.70+DX+IC,2)
   N=MOD(N,10)+1 $ R=R+.1 $ IF(R.LT.RL) GO TO 10
   A=ANG-(IC+1)*45. $ DX=10.*DX $ DY=10.*DY
   C=-.175+.125*IC $ D=.19+.35*IC
   X=X0+C*DX-D*DY $ Y=Y0+C*DY+D*DX
   R=AMAX1(ARS(PMIN),ARS(RMIN+DR*RL)) $ R=ALOG10(R)
   IR=INT(ARS(R)) $ IF(R.LT.0.) IR=- (IP+1) $ IR=IR-MOD(IR,3)
   P1=RMIN/10.**IR $ DP1=DP/10.**IR $ P=0.
   FCODE(7,101,S)R1 $ CALL SYMBOL(X,Y,.07,S,A,7) $ R1=R1+DR1
   X=X+DX $ Y=Y+DY $ R=R+1. $ IF(R.LE.PL) GO TO 20
   R=(RL-.1*NNC)/2. $ C=.1+.5*IC
   X=X0+R*DX-C*DY $ Y=Y0+R*DY+C*DX
   CALL SYMBOL(X,Y,.1,L,ANG,NNC) $ IF(IR.EQ.0) RETURN
   ENCODE(5,102,S) $ CALL SYMBOL(399.,999.,.10,S,ANG,5)
   CALL WHEREF(X,Y,A)
   ENCODE(3,103,S) IR $ CALL SYMBOL(X,Y,.07,S,ANG,3)
   FORMAT(57,2)
   FORMAT(5H +10)
   FORMAT(I3)
   RETURN $ ENN
```

```
SUBROUTINE SCALE(DATA,LENGTH,N,K)
REAL DATA(N),LENGTH,SF(5)
DATA SF/1.0,2.0,2.5,5.0,10.0/
OMIN=DMAX=DATA(1)
DO 10 I=1,N
IF (DATA(I).LT.OMIN) OMIN=DATA(I)
IF (DATA(I).GT.DMAX) DMAX=DATA(I)
DATA(N+1)=OMIN+DATA(N+2)=1.0
IF (LENGTH.LE.0.0)OP.DMAX.EQ.OMIN) RETURN
RAWSF=(DMAX-OMIN)/LENGTH
SFEXP=AIN(TALOG10(RAWSF))
IF (RAWSF.LT.1.0) SFEXP=SFEXP-1.0
SFMANT=RAWSF*10.0*(-SFEXP)
DO 20 I=1,5
IF (SF(I).GT.SFMANT) GO TO 30
PRINT*, " SCALE: SCALE FACTOR ERROR. " $ RETURN
SFVICE=SF(I)*10.0**SFEXP
ADJMIN=AIN(T(OMIN/SFVICE))*SFVICE
IF (ADJMIN.GT.OMIN) ADJMIN=ADJMIN-SFVICE
IF ((OMAX-ADJMIN)/SFVICE.LT.LENGTH) GO TO 40
IF (I.LT.5) SFVICE=SF(I+1)*10.0**SFEXP
IF (I.EQ.5) SFVICE=20.0*10.0**SFEXP
ADJMIN=AIN(T(OMIN/SFVICE))*SFVICE
IF (ADJMIN.GT.OMIN) ADJMIN=ADJMIN-SFVICE
DATA(N+1)=ADJMIN $ DATA(N+2)=SFVICE $ RETURN $ END
10
20
30
40
```



## Appendix E

### SBC 80/20 Hardware Description (Ref 16)

#### Memory

There are two types of memory on the SBC 80/20 board: random access memory (RAM) and read-only memory (ROM). Eight INTEL 2113 static RAM devices provide 2048 (2K) X 8-bits of read/write storage. The RAM address space is located from 3800H to 3FFFH by jumper connection 120-121. Four Intel 2708 Erasable and Electrically Programmable Read Only Memory chips provide 4096 (4K) X 8-bits of ROM. The ROM address space is located from 0000H to 0FFFH. A complete memory map is given in Table III, Appendix A. (The functional characteristics of the memories are given in the SBC 80/20 Hardware Reference Manual (Ref 16: Ch 3, 21-29).)

#### Parallel I/O Interface

Two 8255 Programmable Peripheral Interfaces provide the input ports from the DAS 1128 and the heart rate detector. The remainder of this section describes these interfaces as configured for this specific application. A complete operational summary of the 8255 is available in the hardware reference manual (Ref 16: Ch 3, 51-73).

The 8255 contains three 8-bit ports (A, B, and C). The operating system configures these ports to the strobed input mode (mode 1, control word 'B6H'). This configuration provides for two input ports (A and B) and a control port (C). Each input port contains an input latch to hold the received data while the control port consists of six control bits and two output bits. (The SBC 80/20 modifications

listed in the hardware reference manual (Ref 16: Ch 4, 17-18 and 28-29) were accomplished.)

8255 #1 interfaces the DAS 1128 to the SBC 80/20. Port A receives the 8 most significant bits (MSB); Port B receives the 4 least significant bits (LSB) and the 4 multiplexer address bits. The Port C control bits are used as follows:

- $C_0$ :  $INTR_B$  - interrupt request (not used)
- $C_1$ :  $IBF_B$  - "high" indicates the data has been loaded into the input latch
- $C_2$ :  $\overline{STB}_B$  - "low" loads the data into the input latch
- $C_3$ :  $INTR_A$  - (not used)
- $C_4$ :  $\overline{STB}_A$  - (same as above)
- $C_5$ :  $IBF_A$  - (same as above)
- $C_6$ : (output) - set by control word ' $\emptyset DH$ '  
reset by control word ' $\emptyset CH$ '
- $C_7$ : (output) - set by control word ' $\emptyset FH$ '  
reset by control word ' $\emptyset EH$ '

The  $\overline{EOC}$  signal from the DAS 1128 provides the  $\overline{STB}_A$  and  $\overline{STB}_B$  signals. Bit  $C_6$  provides the  $\overline{STROBE}$  signal to the DAS 1128; while bit  $C_7$  provides the  $\overline{TRIG}$  signal (see Appendix F). The complete pin assignments for this interface are given in Table IV, Appendix A.

8255 #2 interfaces the heart rate detector to the SBC 80/20. Port A receives the heart rate count; Port B is not used. The Port C control bits are used as follows:

- $C_0 - C_2$ : (not used)
- $C_3$ :  $INTR_A$  - (not used)

$C_4$ :  $\overline{STB}_A$   
 $C_5$ :  $IBF_A$   
 $C_6 - C_7$ : (not used)

A strobe signal ( $\overline{STB}$ ) is generated by the heart rate detector when the count is completed (see Appendix G). The complete pin assignments for this interface are given in Table V, Appendix A.

The 8255 I/O port addresses are given in Table VIII; a complete I/O port addressing table is given in the hardware reference manual (Ref 16: Ch 2, 7).

Table VIII  
8255 I/O Port Addresses

I/O Device	I/O Port Address (hexadecimal)	
8255 #1	Port A	E4
	Port B	E5
	Port C	E6
	Control	E7
8255 #2	Port A	E8
	Port B	E9
	Port C	EA
	Control	EB

### Serial I/O Interface

The 8251 USART provides the output port to the Hazeltine 2000 video terminal (the interface connector pin assignments are listed in Table VI, Appendix A). The remainder of this section describes this interface as configured for the IFPDAS II prototype. A complete summary of the 8251 is available in the hardware reference manual (Ref 16: Ch 3, 34-51).

The system software configures the 8251 as an asynchronous receiver/transmitter. A '4EH' mode instruction programs the USART to the asynchronous mode with 1 stop bit, no parity check, 8 transmitted bits, and a baud rate factor of 16X. A '37H' command instruction sets the request-to-send and data-terminal-ready signals high, enables the receive and transmit capabilities, and resets the error flags. Interval timer '2' supplies the baud rate clock (see the following section).

### Interval Timers

The 8253 Programmable Interval Timer includes three separate counters (0, 1, and 2). Counter 0 is used as a frequency divider to produce the 225 Hz clock required by the heart rate circuit (see Appendix G). Control word '36H' configures counter 0 as a square wave rate generator. The counter is then loaded with 12ABH (4779) which produces the desired frequency. Counter 1 is used as a real time clock to inform the CPU of every 50 msec interval. Control word '70H' configures counter 1 to interrupt the CPU when the count is complete. The counter is loaded with D1E5H (53,733) which is equivalent to 49.97 msec. The time required for the CPU to handle the interrupt and reset the timer brings the total time between interrupts to 50.00 msec.

Counter 2 is used as a frequency divider to produce the baud rate clock required by the 8251 USART. Control word 'B6H' configures counter 2 as a square wave rate generator. The counter is then loaded with 0038H (56) which produces the desired frequency. A complete summary of the 8253 is contained in the hardware reference manual (Ref 16: Ch 3, 73-87).

### Interrupt Controller

The 8259 Programmable Interrupt Controller provides the capability to recognize interrupt requests, and based on that request, to jump to any location in the memory map. This section describes the operation of the 8259 in the IFPDAS II prototype. A complete operational summary of the 8259 is given in the hardware reference manual (Ref 16: Ch 3, 87-110).

The 8259 uses a jump table stored in PROM (03E0H to 03FFH) to pass control to the interrupt handling routine. When the 50 msec timer expires, an interrupt request (IR3) is sent by the timer to the 8259. The 8259 accepts this request and sends an interrupt to the 8080 CPU. After the CPU acknowledges the request, the 8259 "calls" the fourth entry of the jump table which causes a branch to the 50 msec timer interrupt handler (T50MS).

The operating system programs the 8259, during the power-up routine, to accomplish this task. Two initialization command words (ICW) are required to inform the 8259 of the location and length of the jump table. ICW1 ('F6H') and ICW2 ('03H') "tell" the 8259 that the jump table starts at 03E0H and that the call address interval is 4. After the 8259 receives these two words, it is in the normal (fully nested) mode and is ready to operate.

The 8259 is programmed by the CPU to ignore IR2. IR2 is generated by counter 0 which has a special function when the SBC 80/20 monitor is executing (Ref 15:8). Since counter 0 is used for a different purpose in the prototype than in the monitor, the 8080 sends an operational command word (OCW) to the 8259. OCW1 (= '04H') masks off IR2 so that this request is never "seen" by the 8259.

The final command sent to the 8259 by the operating system is an end of interrupt (EOI) command word (OCW2 = '20H'). This command resets the in-service bit (IS3) which allows IR3 to request another interrupt.

## Appendix F

### DAS 1128 Hardware Description (Ref 17)

The DAS 1128 is a complete self-contained miniature high speed data acquisition system which is described in the attached 8-page pamphlet. The module is hard-wired to its IFPDAS II prototype configuration; these modifications are listed in Table IX along with their function. These hard-wired modifications are also reflected in Figure 17, Appendix A. The complete operating characteristics are given in the pamphlet.

Table IX

DAS 1128 Hardwired Modifications

Function	Jumper Connection
16 single-ended inputs	11B to 11T 12B to 2B 17B to 19T 18T to 18B
Full range scale 0 - 5.12 volts	12T to 13B 14T and 14B to 13T 15B to 16B
Full 12 bit operation	28T to DIG GND
Output code: Unipolar Binary	17T to -15 volts 29T (B1) is MSB
Sequentially triggered multiplexer addressing	24B to +5 volts STROBE to 8255 #1 TRIG to 8255 #1
Sequence 0 to 6, then repeat	4 OUT and 2 OUT to external NAND gate Output of NAND gate to 25B
Highest accuracy	CLK TRIM to DIG GND (provides 2.08 microsec/ bit conversion time) DLY TRIM to DIG GND ±15 V return to ANA RTN +5 V return to DIG RTN



### FEATURES

- Complete Data Acquisition System
- 12 Bit Digital Output
- 16 Single or 8 Differential Analog Inputs
- High Throughput Rate
- Selectable Analog Input Ranges
- Versatile Input/Output/Control Format
- Low 3 Watt Power Dissipation
- Small 3" x 4.6" x 0.375" Module



### GENERAL DESCRIPTION

The DAS1128 is a complete self-contained miniature high speed data acquisition system. The compact 3" x 4.6" x 0.375" module provides the designer with an easily implemented solution to the data acquisition problem. It contains an analog input signal multiplexer, a sample-and-hold amplifier, a 12 bit A/D converter, and all of the programming, timing and control circuitry needed to perform the complete data acquisition function.

The DAS1128 is a high performance device which can digitize an analog signal to an accuracy of  $\pm 1/2$  LSB out of 12 bits, relative to full scale. It has  $\pm 8$  ppm/ $^{\circ}$ C gain temperature coefficient, and the maximum throughput rate can be varied from 50,000 conversions/second for a 12 bit conversion from different analog input channels, to 200,000 conversions/second for a successive 4 bit conversion made on a single channel.

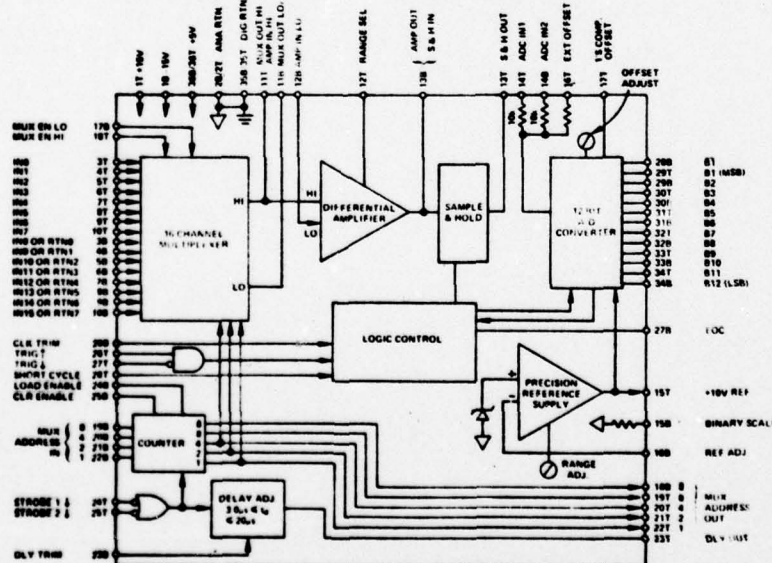


Figure 1. Functional Block Diagram

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 Mid-West 312/894-3300  
 Texas 214/231-5094  
 TWX: 710/394-6577

# SPECIFICATIONS (typical @ +25°C and ±15V unless otherwise noted)

## ANALOG INPUTS

Number of Inputs to Multiplexer	16 Single Ended, 8 True-Differential, 16 Pseudo-Differential
Input Voltage (Full Scale Range)	-10V to +10V, 0V to +10V, -5V to +5V, 0V to +5V, -10.24V to +10.24V, 0V to +10.24V, -5.12V to +5.12V, or 0V to +5.12V.
Maximum Input Voltage	±15V
Input Current (per channel)	5nA max
Input Impedance	>10 <sup>10</sup> ohms
Input Capacitance	10pF for "OFF" channel 100pF for "ON" channel
Input Fault Current (power off or MUX failure)	Internally limited to 20mA
Direct ADC Input Impedance	10kΩ for each input line

## ACCURACY<sup>1</sup>

Resolution	12 Bits
Error Relative to F.S.	±½LSB
Quantization Error	±½LSB
Differential Nonlinearity Error	
● 33kHz throughput rate	±½LSB, 1LSB max
● 50kHz throughput rate	±1LSB
Noise Error	±½LSB
-FS to +FS Error Between Successive Channel Transitions	±1LSB

## TEMP. COEFFICIENTS

Gain	8ppm/°C, 20ppm/°C max
Offset	5ppm/°C, 15ppm/°C max
Differential Nonlinearity	2.5ppm/°C, 6ppm/°C max

## SIGNAL DYNAMICS

Throughput Rate (12 Bits)	50kHz (max) (includes 5μsecs for MUX and SHA settling time plus 15μsecs for ADC)
MUX Crosstalk ("OFF" channels to "ON" channel)	>80dB down @ 1kHz
Differential Amplifier CMRR	70dB to 1kHz
SHA Acquisition Time to 0.01%	4.5μsec max
SHA Aperture Uncertainty	10nsec
SHA Feedthrough	70dB down @ 1kHz

## DIGITAL INPUT SIGNALS

Compatibility	Standard DTL/TTL logic levels, 1 unit load/line
MUX Address Inputs (8, 4, 2, 1; Pins 19B through 22B)	Positive true natural binary coding selects channel for random addressing mode. Must be stable for 100nsec after STROBE.
MUX ENABLE HI (Pin 18T)	High (logic "1") input enables MUX "HI" output (for inputs 0 through 7)
MUX ENABLE LO (Pin 17B)	High (logic "1") input enables MUX "LO" output (for inputs 8 through 15)
STROBE (Pin 24T or 25T)	Negative going transition (logic "1" to logic "0") updates MUX address register. STROBE 1 must be a logic "1" to enable STROBE 2. STROBE 2 must be at logic "1" to enable STROBE 1.
LOAD ENABLE (Pin 24B)	High (logic "1") input allows next STROBE command to sequentially advance MUX address register. Low (logic "0") input allows next STROBE command to update MUX address register according to external address inputs.
CLEAR ENABLE (Pin 25B)	Low (logic "0") input allows next STROBE command to reset MUX address to channel "0" overriding LOAD ENABLE.
TRIGGER (Pin 26T)	Positive going transition (logic "0" to logic "1") initiates A/D conversion (even during conversion); TRIGGER (Pin 27T) must be at logic "0" to allow TRIGGER function.
TRIGGER (Pin 27T)	Negative going transition (logic "1" to logic "0") initiates A/D conversion; Pin 26T (TRIGGER) must be at logic "1" to allow TRIGGER function.

<sup>1</sup> Warmup time to rated accuracy is 5 minutes.

<sup>2</sup> Specification applies only when tracking +15V and -15V supplies are used, and for slowly occurring variations in power supply voltages.

Specifications subject to change without notice.

## DIGITAL OUTPUT SIGNALS

Compatibility	Standard DTL/TTL logic levels; 5 unit loads/line.
Parallel Outputs	B7, B1 through B12
Coding	Natural binary, two's complement, offset binary, or one's complement. Pin selectable.
MUX Address Outputs (8, 4, 2, 1; pins 18B, 19T through 22T)	Positive true natural binary coding indicates channel selected.
DELAY OUT (Pin 23T)	Negative going transition (logic "1" to logic "0") occurring normally 5μsecs (adjustable from 3.0μsecs to 20μsecs) after STROBE command initiates A/D conversion automatically when connected to the TRIGGER.
EOC (Pin 27B)	High (logic "1") output during A/D conversion.

## ADJUSTMENTS & TRIMS

Offset Adjust	
Internal Adjustment (Externally Accessible)	±10LSB's (min)
Remote External Adjustment (Pin 16T)	±10LSB's (min)
Range Adjust	
Internal Adjustment (Externally Accessible)	±10LSB's (min)
Remote External Adjustment (Pin 16B)	±10LSB's (min)
Clock Trim (Pin 26B)	
Factory Setting (Pin 26B "OPEN")	1.25μs/Bit
External Adjustment Range	1.25μs/Bit to 2.08μs/Bit
Delay Trim (Pin 23B)	
Factory Setting (Pin 23B "OPEN")	3.0μs
External Adjustment Range	3.0μs to 20μs

## CONTROLS

SHORT CYCLE (Pin 28T)	Connect to ground for full 12 bit resolution. Connect to B <sub>n</sub> output for resolution to B <sub>n-1</sub> bits. Random, sequential continuous, and sequential triggered. Pin selectable.
Channel Selection Mode (MUX Address Loading Mode)	Normal (input channel remains selected during its A/D conversion) and overlap (next channel selected during A/D conversion). Pin selectable.
A-D Conversion/Channel-Select Sequences	Differential Amplifier gain control: connect to ANA RTN (Pin 2T) for X1 gain; connect to AMP OUT (Pin 13B) for X2 gain. This control is used in FSR selection procedure. Connect to REF ADJ (Pin 16B) to set reference to 10.24V. This control is used in FSR selection procedure, see Table II.
Range Select (Pin 12T)	Ground for 1's complement output code; connect to -15VDC for other available codes.
BINARY SCALE (Pin 15B)	
OUTPUT CODING (Pin 17T)	

## POWER REQUIREMENTS

+15V ±3%	40mA, 50mA max
-15V ±3%	70mA, 100mA max
+5V ±5%	250mA, 500mA max
Power Supply Sensitivity <sup>2</sup> :	
Gain	±2.0mV/V
Offset	±4.0mV/V
Ref	±0.5mV/V

## ENVIRONMENT & PHYSICAL

Operating Temperature	0° to +70°C
Storage Temperature	-25°C to +85°C
Relative Humidity	Up to 95% non-condensing
Electrical Shielding	RFI & EMI 6 sides (except connector area)
Packaging	Insulated steel cased module 3.00" x 4.60" x 0.375"

## PRICE

\$295.00 (1-9), price includes mating right-angle connector.

## THEORY OF OPERATION

A block diagram of the DAS1128 is shown in Figure 1. Analog input signals are applied to the various inputs of the 16 channel CMOS multiplexer. This multiplexer in conjunction with the differential amplifier that follows it, can be configured by the user to accept 16 single ended analog inputs, or 8 fully differential analog inputs. It can also be connected as a 16 channel "pseudo-differential" input device, which permits some of the benefits of differential operation while maintaining a 16 channel input capability.

The differential buffer amplifier is gain programmable by the user via jumpers at the module pins. This feature, along with the selectable reference voltages, permits the user to set up the DAS1128 to operate on any of 8 input voltage ranges. The differential amplifier drives a sample-and-hold amplifier, whose function it is to hold the selected analog input signal at a constant level while the A/D converter is making a conversion.

The A/D converter is a high speed 12 bit successive approximation device that has been designed using the Analog Devices' AD562, 12 bit integrated circuit D/A. The reference voltage for the conversion is supplied by an adjustable precision reference circuit that has a temperature coefficient of 5ppm/°C.

In addition to these basic functional blocks, the DAS1128 also contains all of the clock circuitry necessary to perform the complete data acquisition function. The internal clock can be externally adjusted to provide various throughput rates at different accuracies. Input channel addressing logic is provided, as is the capability to short cycle the A/D converter (i.e. perform conversions of less than 12 bits resolution). It is also possible for the user to adjust the time interval between input channel selection and the commencement of a conversion. The user can thus trade off speed vs. accuracy in the settling time of the multiplexer and sample-and-hold amplifier, as well as speed versus accuracy of the A/D converter.

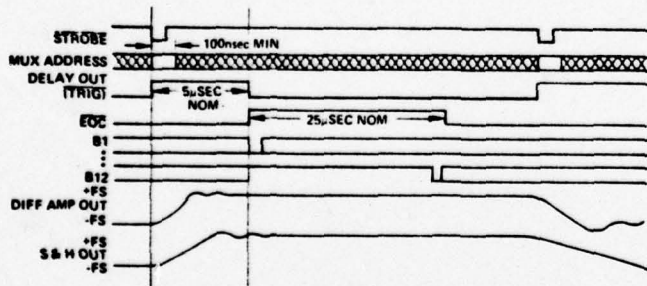


Figure 2. Simplified Timing Diagram, Showing Time-Interval Assignments and Constants.

## INPUT CONNECTIONS

As shown in Figure 3, three input configurations can be used. 16 single-ended inputs (3a) can be connected to the multiplexer, all referenced to analog gnd. In the second configuration (3b), the inputs are connected individually as 8 true differential pairs. In this case the differential amplifier is connected "Differentially" with the output of the MUX. Finally, a "Quasi-Differential" connection (3c) can be realized under favorable ground path conditions. In this configuration the differential amplifier Lo terminal is used as the ground return

for all sensors. In each of these input schemes, it should be noted that the input multiplexer has been designed to protect itself and signal sources from both overvoltage failure and from fault currents due to power-off loading or MUX failure.

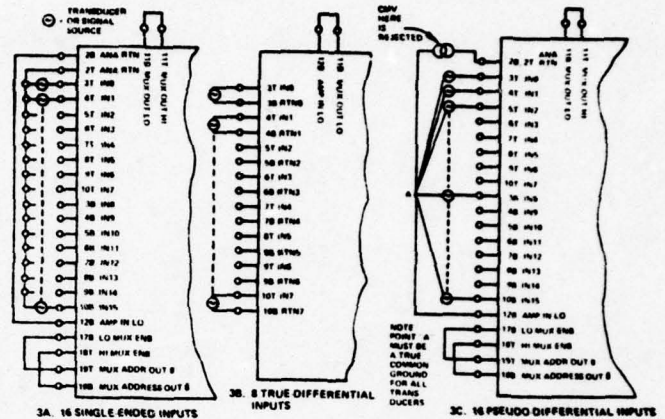


Figure 3. Signal Input Connections for Three Different Configurations.

Full scale range of the DAS1128 may be set by appropriate jumper connections for 8 different ranges: 0 to +10V; 0 to +5V; 0 to +10.24V; 0 to +5.12V; -10 to +10V; -5 to +5V; -10.24 to +10.24V; -5.12 to +5.12V.

Note that 10.24 and 5.12 ranges are commonly used since conversion increments become 5mV/bit, 2.5mV/bit, and 1.25mV/bit.

## MUX AND S/H DYNAMICS - OVERLAP MODE

The overlap mode is defined as the ability of MUX to accept a new channel address thereby selecting the next channel to be sampled while the previously acquired sample is being held by the S/H for conversion. The dynamic characteristics of the S/H circuit are shown in Figure 4. Maximum throughput rates are obtainable when a single channel is held at a single address and the channel is sampled repeatedly. In a dynamic condition, data-throughput rates obtainable are shown in Figure 5.

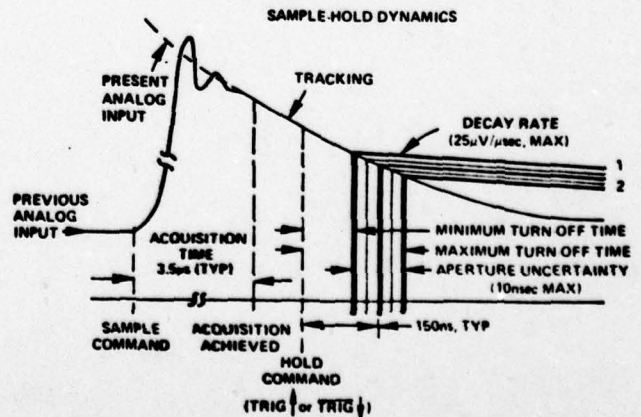


Figure 4. Sample-Hold Parameters Defined and Specified

### SHORT CYCLE

It is possible to short cycle the DAS1128, i.e. stop the conversion after less than 12 bits. This can be done by connecting an external jumper between short cycle terminal and one of the output terminals. With shorter cycles the attainable throughput rate increases, see Figure 5. In short cycle operation the  $\Delta C$  will decrease proportionately to the number of bits selected. Note the short cycle terminal *must* be grounded for full 12-bit operation.

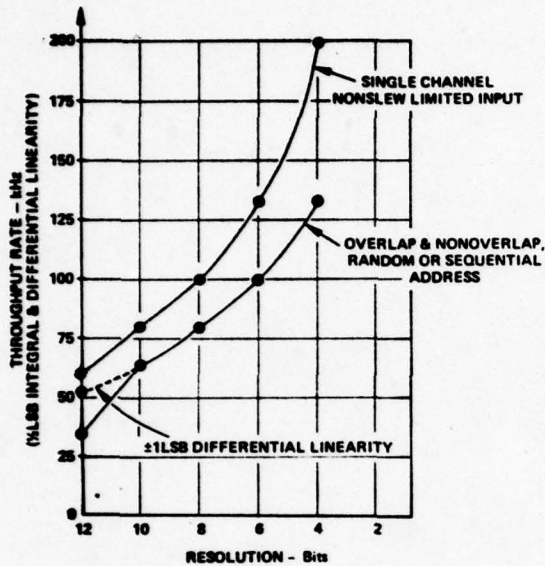


Figure 5. DAS1128 Throughput Rates

### MUX ADDRESSING

External terminals have been provided for the address counter. Thus the address counter can be configured to produce the following modes: Continuous sequential scanning (free running), sequential scanning with external step command, abbreviated scan continuously, random channel selection. See Figure 6 and set up procedure for details.

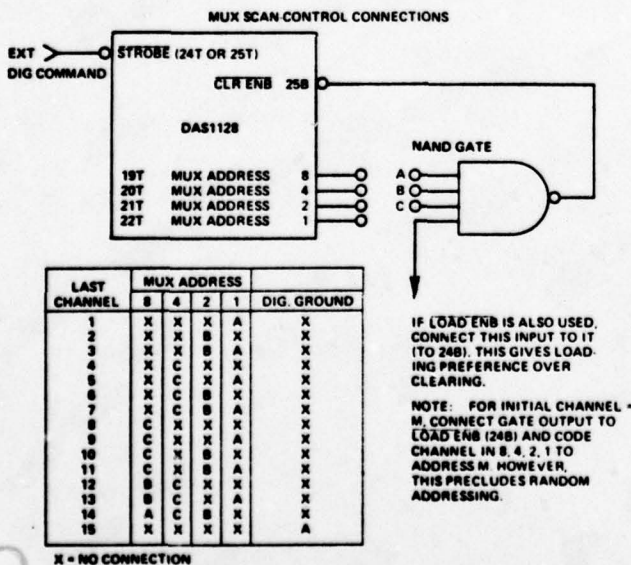


Figure 6. To shorten scanning sequency of multiplexer channels, make the appropriate connections, (as shown in the chart) between an external NAND gate and MUX ADDRESS terminals 197 to 217.

### GROUNDING CONSIDERATIONS

Attention should be given to the methods of connection for electrical returns and voltage reference points. Analog return (ANA RTN) and digital return (DIG RTN) are provided. The following rules should be applied when integrating the DAS1128 into the system.

1. If the  $\pm 15V$  power supply is floating (for optimum analog accuracy), connect its return to ANA RTN (Pin 2B or 2T). If the  $\pm 15V$  power supply is *not* floating, connect its return to DIG RTN (Pin 35T or 36T).
2. Connect the +5V supply return to DIG RTN (Pin 35T or 36T). If this supply also powers additional equipment, run separate, parallel returns to the equipment ground and to DIG RTN (Pin 35T or 36T).
3. To minimize signal grounding problems, single-ended input signals should only be returned to ANA RTN (Pin 2B or 2T). If this is not possible, then connect the input signals in either the "true differential" or "pseudo-differential" configurations (see Figure 3).
4. Connect computer ground to DIG RTN (Pin 35T or 35B). Use heavy wire or ground planes.
5. The computer chassis should be connected to the computer and power supply grounds at only one point.
6. Connect the third-wire ground from main AC power input to the computer power supply return.

### GAIN AND OFFSET ADJUSTMENTS

The DAS1128 is calibrated with external gain and offset adjustment potentiometers connected as shown in Figure 7 and 8. The offset adjustment potentiometer has an adjustment range of at least  $\pm 10LSB$ 's, and the gain range adjustment potentiometer has an adjustment range of at least  $\pm 10LSB$ 's.

Offset calibration is not affected by changes in gain calibration, and should therefore be performed prior to gain calibration. Proper gain and offset calibration requires great care and the use of extremely sensitive and accurate reference instruments. The voltage standard used as a signal source must be very stable. It should be capable of being set to within  $\pm 1/10LSB$  of the desired value at any point within its range.

These adjustments are not made with zero and full scale input signals, and it may be helpful to understand why. An A/D converter will produce a given digital word output for a small range of input signals, the nominal width of the range being one LSB. If the input test signal is set to a value which should cause the converter to be on the verge of switching between two adjacent digital outputs, the unit can be calibrated so that it does switch at just that point. With a high speed convert command rate and a visual display, these adjustments can be performed in a very accurate and sensitive way. *Analog Devices' Conversion Handbook* gives more detailed information on testing and calibrating A/D converters.

### OFFSET CALIBRATION

For unipolar operation set the input voltage precisely to +0.0012V and adjust the offset potentiometer until the converter is just on the verge of switching from 000000000000 to 000000000001.

For  $\pm 5V$  bipolar operation set the input voltage precisely to -4.9988V; for  $\pm 10V$  units set it to -9.9976V. Adjust the offset

potentiometer, Figure 7, until Offset Binary coded units are just on the verge of switching from 000000000000 to 000000000001 and Two's Complement coded units are just on the verge of switching 100000000000 to 100000000001.

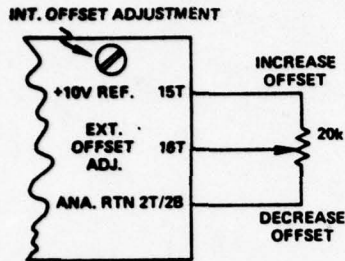
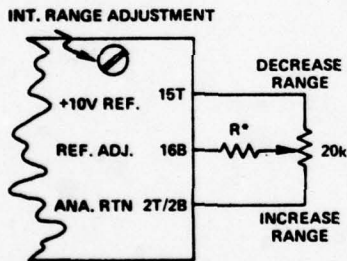


Figure 7. Ext. Offset Adjustment

### GAIN CALIBRATION

Set the input voltage precisely to +9.9963V for unipolar operation, +4.9963V for inputs of  $\pm 5V$  or +9.9926V for inputs of  $\pm 10V$ . Note that these values are  $1/2$ LSB's less than nominal full scale. Adjust the 20k variable gain resistor, Figure 8, until Binary and Offset Binary coded units are just on the verge of switching from 11111111110 to 11111111111 and Two's Complement coded units are just on the verge of switching from 01111111110 to 01111111111.



\*R SHOULD BE AT LEAST 47k $\Omega$  AND NO MORE THAN 470k $\Omega$ . REDUCING R INCREASES ADJUSTMENT RANGE.

Figure 8. Ext. Ref. Adjustment

### CLOCK RATE ADJUSTMENT

The clock rate may be adjusted for best conversion time/accuracy trade-off. The conversion time is varied by means of the external circuitry shown in Figure 9. An open CLK TRIM terminal (Pin 26B) results in 1.25 $\mu$ sec/bit nominal conversion time. A grounded CLK TRIM terminal (for highest accuracy) results in 2.08 $\mu$ sec/bit conversion.

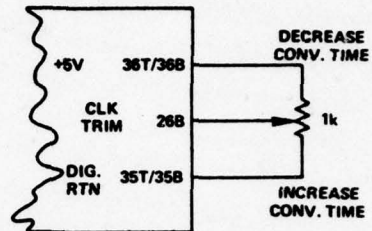


Figure 9. Clock Trim

### DELAY TIME ADJUSTMENT

The DLY OUT signal may be adjusted to vary the A/D converter triggering time by means of the external circuitry shown in Figure 10. An open DLY TRIM terminal (Pin 23B) results in a nominal delay time of 3.0 $\mu$ sec. A grounded DLY TRIM terminal (for highest-accuracy) results in 20 $\mu$ sec delay time nominal.

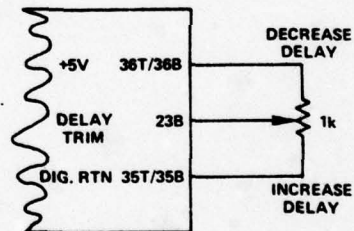


Figure 10. Delay Trim

TABLE I

INPUT CONFIGURATION	ANALOG INPUT CONNECTIONS	ANALOG INPUT RETURN	JUMPER CONNECTIONS
16 Single-Ended Inputs (Figure 3a)	3T thru 10T and 3B thru 10B	All input returns to 2B or 2T	11B to 11T 12B to 2B or 2T 17B to 19T 18T to 18T
8 Differential Inputs (Figure 3b)	3T thru 10T	3B thru 10B	11B to 12B 17B to 18T to "1"
16 Pseudo-Differential Inputs (Figure 3c)	3T thru 10T and 3B thru 10B	Common Input return to 12B	11B to 11T 17B to 19T 18T to 18B

## RECOMMENDED SET-UP PROCEDURE

### 1. Select input configuration, see Table I.

### 2. Select MUX address mode.

The method of addressing the multiplexer can be selected by connecting the unit as follows:

**RANDOM.** Set Pin 24B (LOAD ENB) to logic "0". The next falling edge of STROBE will load the address presented to Pins 19B through 22B (8, 4, 2, 1). The code on these lines must be stable during the falling edge of STROBE plus 100nsec.

**SEQUENTIAL FREE RUNNING.** Set to logic "1", Pin 24B (LOAD ENB) and 25B (CLR ENB). Connect Pin 27B (EOC) to Pin 24T (STROBE T). Connect Pin 23T (DLY OUT) to Pin 27T (TRIG). Use Pin 26T (TRIG) as a run/stop control (i.e. A/D conversion will continue while TRIG is high and will stop while TRIG is low).

**SEQUENTIAL TRIGGERED.** Set to logic "1", Pins 24B (LOAD ENB) and 25B (CLR ENB). Connect Pin 24T (STROBE) to external triggering source. The multiplexer address register will automatically advance by one channel whenever a STROBE command is received. The initial channel can be selected by setting Pin 24B (LOAD ENB) to logic "0" during only one STROBE command. The multiplexer address will then be determined by the logic levels on Pins 19B through 22B (the external MUX address lines). Channel "0" can be selected as the initial channel by setting Pin 25B (CLR ENB) to logic "0" during only one STROBE command. The final channel can be selected by following the procedure presented in Figure 6.

### 3. Select A-D conversion/channel select sequence (see Figure 5).

- (1) **NORMAL** (input channel remains selected during its A/D conversion). Connect Pin 23T (DLY OUT) to Pin 27T (TRIG).
- (2) **OVERLAP** (next channel is selected during A/D conversion). Connect Pin 27B (EOC) to TTL compatible inverter input. Connect inverter output to Pin 24T (STROBE). Connect Pin 23T (DLY OUT) to Pin 27T (TRIG). Adjust the delay to at least 4μsec greater than EOC, 20μsec max (see Figure 10). The signal on Pin 26T (TRIG) serves as RUN/STOP control.
- (3) **REPETITIVE SINGLE CHANNEL.** After selecting the input channel to be repetitively sampled (see MUX ADDRESS MODE, above), set Pin 27T (TRIG) to logic "0". Connect Pin 26T (TRIG) to a triggering source. Conversion process is initiated by positive edge of TRIG command.

### 4. Select output resolution.

- a. **Full 12 bit resolution:** connect Pin 28T (SHT CYC) to Pin 35B (DIG RTN).
- b. **Bn (Bn < 12) bit resolution:** connect Pin 28T to the output pin for Bn + 1.

### 5. Select optimum throughput rate.

The system clock frequency and the STROBE to TRIG delay (if used) can be trimmed to optimize the accuracy/throughput rate trade-off. See Figures 9 and 10.

### 6. Select input voltage full scale range. See Table II.

### 7. Select output digital coding. See Table III.

TABLE II

FOR FULL SCALE RANGE OF:	MAKE THE FOLLOWING CONNECTIONS
0 to +10V	12T to 2T; 14T to 14B to ADC Source*.
0 to +10.24V	same as 0 to +10V, plus 15B to 16B.
0 to +5V	12T to 13B; 14T and 14B to ADC Source*.
0 to +5.12V	same as 0 to +5V, plus 15B to 16B
-10V to +10V	12T to 2T; 14T to 15T; and 14B to ADC Source*.
-10.24V to +10.24V	same as -10V to +10V, plus 15B to 16B
-5V to +5V	12T to 13B; 14T to 15T and 14B to ADC Source*.
-5.12V to +5.12V	same as -5V to +5V, plus 15B to 16B.

\*ADC Source is usually Sample and Hold Output (13T), but may be any signal source including Diff. Amp. Output (13B) if Sample and Hold is not desired.

TABLE III

OUTPUT CODE	CONNECTIONS
Unipolar Binary	Connect 17T to -15V Use 29T (B1) for MSB
2's Complement	Connect 17T to -15V Use 28B (B1) for MSB
Offset Binary	Connect 17T to -15V Use 29T (B1) for MSB
1's Complement	Connect 17T to 2B Use 28B (B1) for MSB

# Timing Diagrams

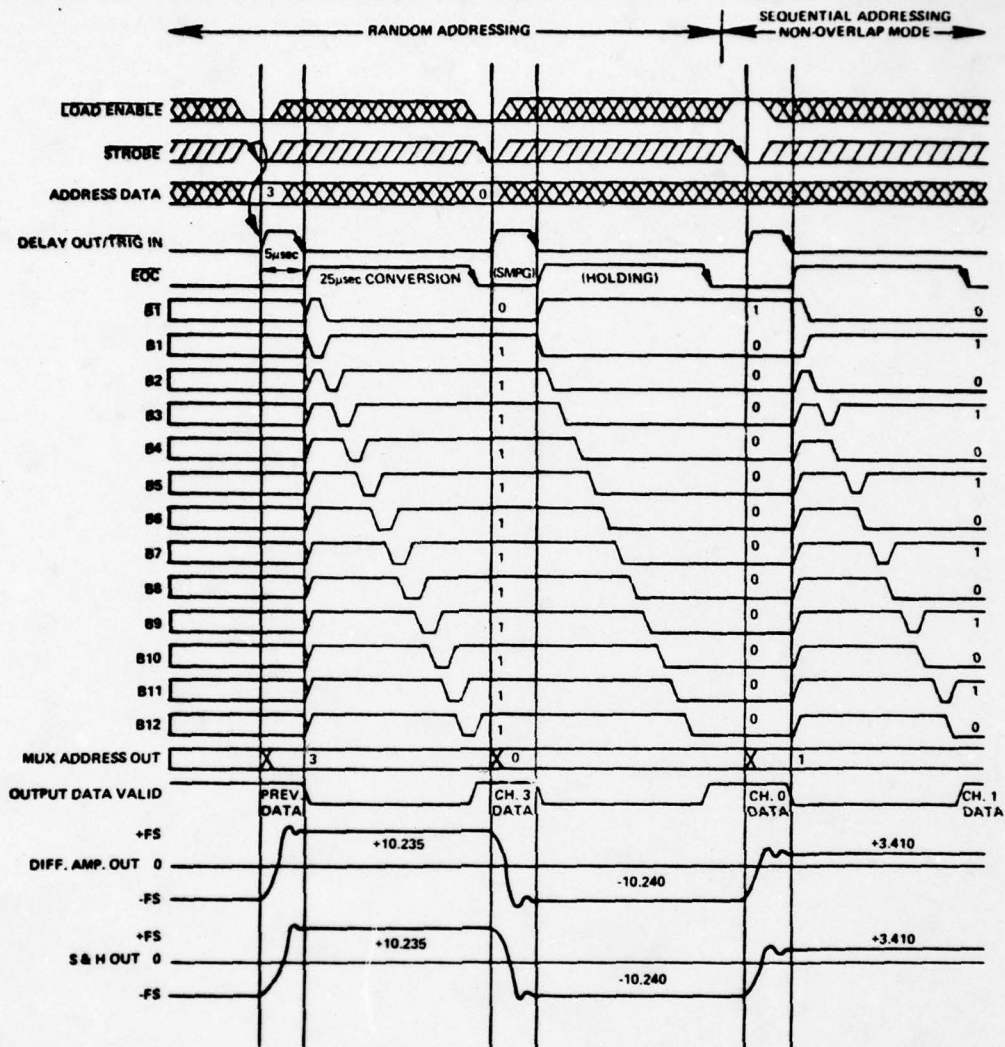


Figure 11. Timing for Non-Overlap Operation in Both Random and Sequential Addressing Modes. For Status Keys and Signal Condition Data, Refer to Box Below.

## SIGNAL CONDITIONS AND STATUS KEYS FOR FIGURES 11 AND 12.

CH. 2 = -3.415V CODE 010 101 010 101  
 CH. 3 = +10.235V CODE 111 111 111 111  
 CH. 0 = -10.240V CODE 000 000 000 000  
 CH. 1 = +3.410V CODE 101 010 101 010

ADC SET UP FOR ±10.24V. INPUT, OFFSET BINARY. (FOR TWO'S COMPLEMENT, USE  $\bar{B}1$  FOR M.S.B.)

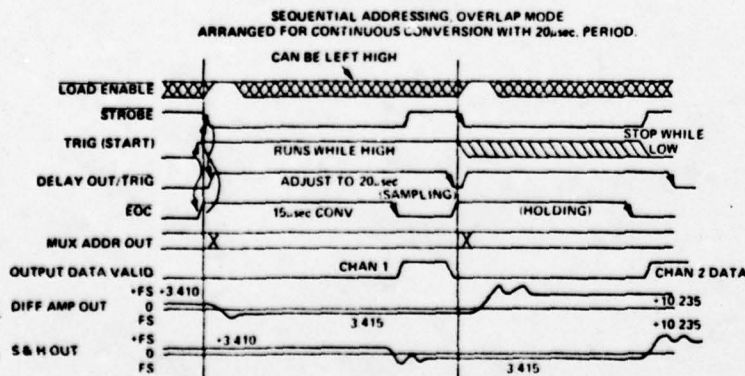


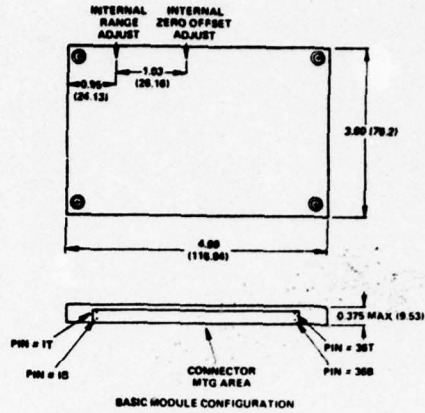
Figure 12. Timing Diagram for Overlap Operation in the Sequential Addressing Mode. For Status Keys and Signal Condition Data, See Box at Right.

KEY	INPUTS	OUTPUTS
XXX	May change	Don't know
ZZZ	May change 0 to 1	Changes 0 to 1
SSS	May change 1 to 0	Changes 1 to 0
OR	Must be stable	Will be stable

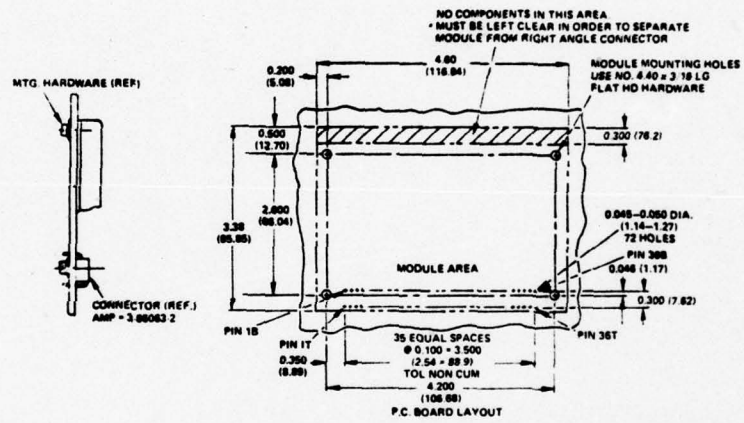
# Outline Drawings and Pin Designations

## DAS1128 Connector Pin Diagram

+15V	1T	1B	-15V
ANA RTN	2T	2B	ANA RTN
CH 0 IN	3T	3B	CH 8 IN (CH 0 RTN)
CH 1 IN	4T	4B	CH 9 IN (CH 1 RTN)
CH 2 IN	5T	5B	CH 10 IN (CH 2 RTN)
CH 3 IN	6T	6B	CH 11 IN (CH 3 RTN)
CH 4 IN	7T	7B	CH 12 IN (CH 4 RTN)
CH 5 IN	8T	8B	CH 13 IN (CH 5 RTN)
CH 6 IN	9T	9B	CH 14 IN (CH 6 RTN)
CH 7 IN	10T	10B	CH 15 IN (CH 7 RTN)
MUX HI OUT	11T	11B	MUX LO OUT
RANGE SEL	12T	12B	AMP IN LO
S & H OUT	13T	13B	AMP OUT
ADC IN 1	14T	14B	ADC IN 2
+10V REF	15T	15B	BINARY SCALE
EXT OFFSET	16T	16B	REF ADJ
OUTPUT CODING	17T	17B	ENABLE LO
ENABLE HI	18T	18B	8 OUT
8 OUT	19T	19B	8 IN
4 OUT	20T	20B	4 IN
2 OUT	21T	21B	2 IN
1 OUT	22T	22B	1 IN
DLY OUT	23T	23B	DLY TRIM
STROBE 1	24T	24B	LOAD ENB
STROBE 2	25T	25B	CLR ENB
TRIG	26T	26B	CLK TRIM
TRIG	27T	27B	EOC
SHT CYC	28T	28B	8T OUT
81 OUT	29T	29B	82 OUT
83 OUT	30T	30B	84 OUT
85 OUT	31T	31B	86 OUT
87 OUT	32T	32B	88 OUT
89 OUT	33T	33B	810 OUT
811 OUT	34T	34B	812 LSB OUT
DIG RTN	35T	35B	DIG RTN
+8V	36T	36B	+5V

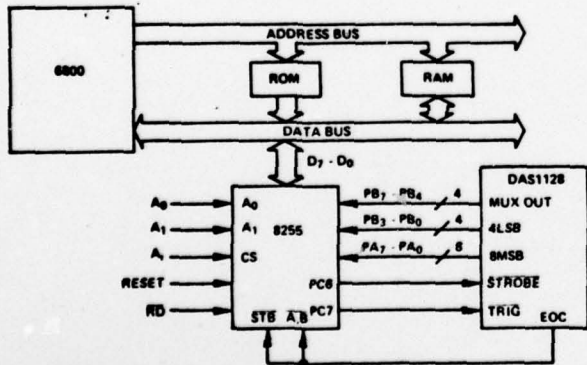


Dimensions shown in inches and (mm).



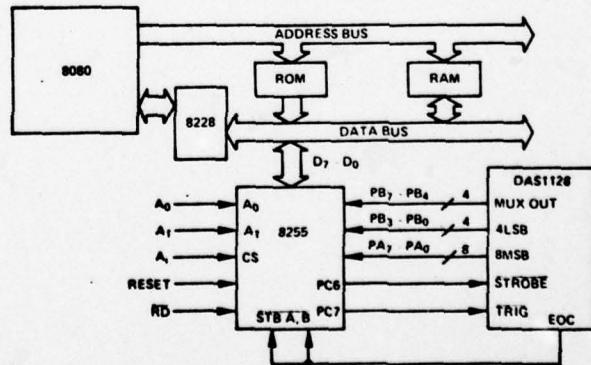
## Typical Applications

### DAS1128 WITH MOTOROLA 6800



- NOTE:
1. 8255 USED IN MODE 1 (STROBED I/O)
  2. PC6 INDEXES MUX TO DESIRED CHANNEL
  3. CS TO A<sub>n</sub> (WHERE A<sub>n</sub> IS AN ADDRESS BIT OTHER THAN A<sub>0</sub> OR A<sub>1</sub>)
  4. PC7 INITIATES CONVERSION
  5. EOC STROBES IN DATA AND MUX INFO
  6. 8255 SHOWN, HOWEVER 6820 CAN ALSO BE USED

### DAS1128 WITH INTEL 8080



- NOTE:
1. 8255 USED IN MODE 1 (STROBED I/O)
  2. CS TO A<sub>n</sub> (WHERE A<sub>n</sub> IS AN ADDRESS BIT OTHER THAN A<sub>0</sub> OR A<sub>1</sub>)
  3. PC6 INDEXES MUX TO DESIRED CHANNEL
  4. PC7 INITIATES CONVERSION
  5. EOC STROBES IN DATA AND MUX INFO



## Appendix G

### Sensor Interfaces

This appendix discusses the three circuits that were designed and built to interface the oxygen partial pressure sensors, the accelerometers, and the heart rate detector to the IFPDAS II prototype.

#### OM11 Sensor Interface (Ref 18)

The circuit in Figure 21 interfaces both Beckman OM11 polarographic sensors to the IFPDAS II prototype. The OM11 sensor is biased to -740 mvolts and produces a current (approximately 2 microAmps for air at room temperature) that is proportional to the oxygen partial pressure of the gas that surrounds the sensor. An amplifier converts this current source to a voltage output. The output signal ranges from 0 to 5.0 volts corresponding to oxygen partial pressures of 0 to 760 mm Hg; the response time for the sensor is 800 msec when exposed to a pure oxygen source.

The  $D_1$ - $R_1$ - $R_2$ - $R_3$  combination acts as a voltage regulator and divider, reducing the -15 volts to the necessary -740 mV sensor bias. The sensor current output is then amplified through amplifier  $A_1$  to produce 1.0 volts (152 mm Hg). The  $C_1$ - $R_5$  combination determines the amplifier gain and provides low pass filtering with a 5 Hz cutoff frequency. Temperature compensation for the sensor is provided by a built-in 10K thermistor ( $R_7$ ) which, in parallel with  $R_8$ , automatically adjusts the gain for changing sensor temperature. Calibration of the

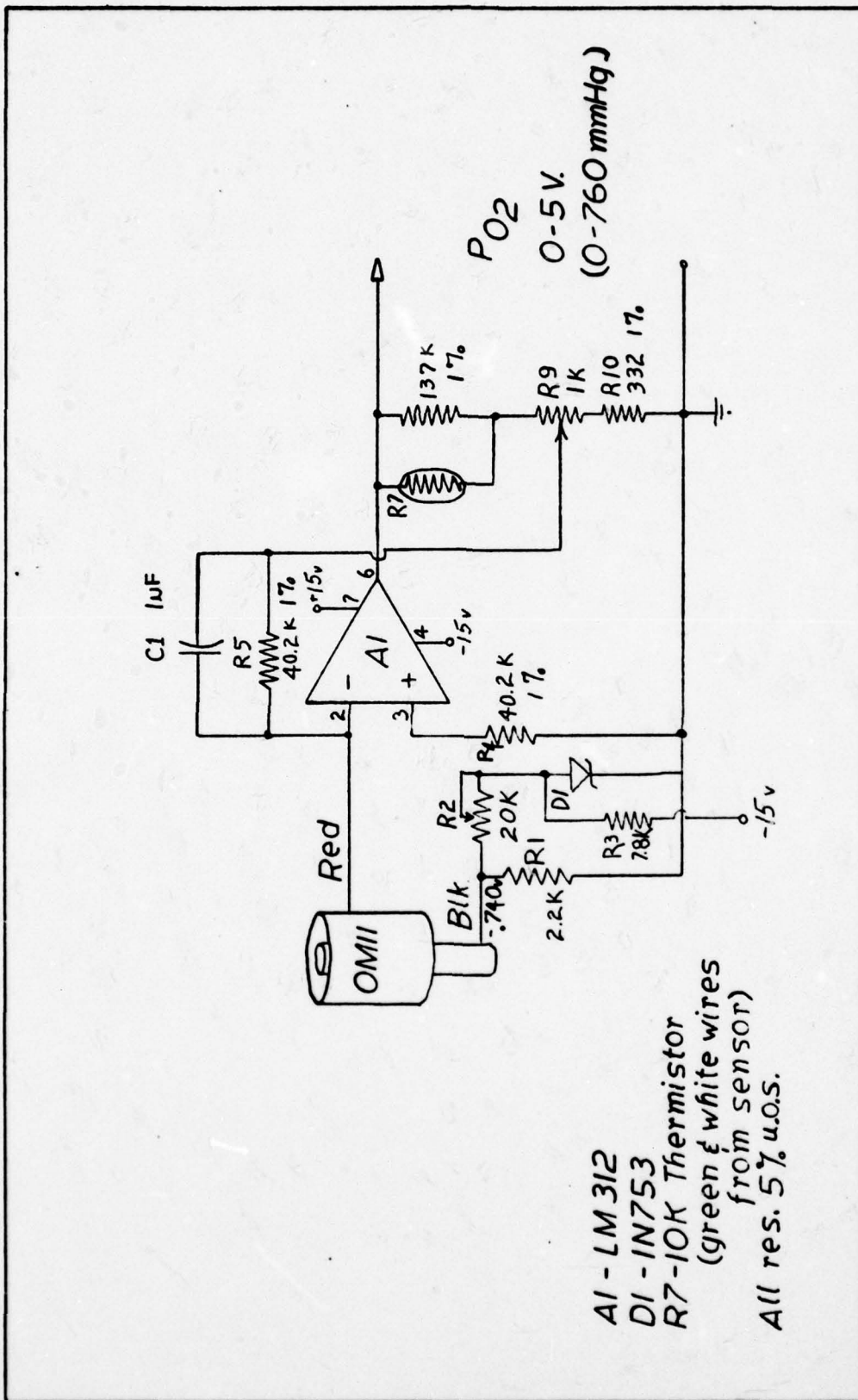


Fig. 21. OM11 Interface Circuit

interface is accomplished using a 100% oxygen source and adjusting potentiometer  $R_9$  for a signal output of 5.0 volts.

If faster sensor response time is necessary, the circuit of Figure 22 can be used to compensate the sensor. This circuit was provided by Beckman Instruments (Ref 19) and it reduces the sensor response time to 100 msec. Power consumption of this circuit can be minimized by utilizing all UA776 amplifiers which can be biased for microwatt power consumption by appropriately selecting the bias resistors (Ref 20: Ch 8, 458-466).

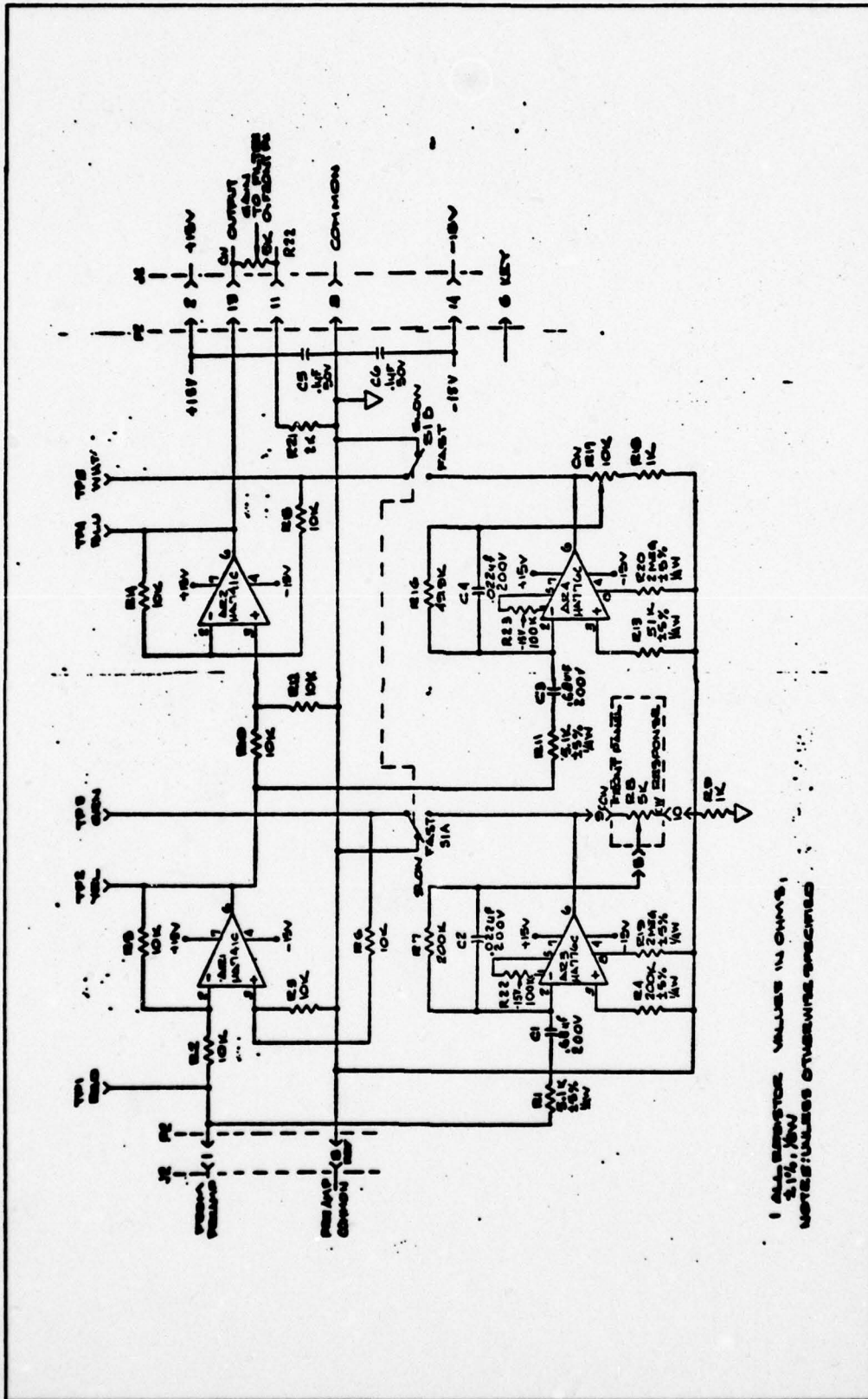


Fig. 22. OM11 Compensator Circuit

## Accelerometer Interfaces

The Statham F-15-340 accelerometer interface circuit is shown in Figure 23.

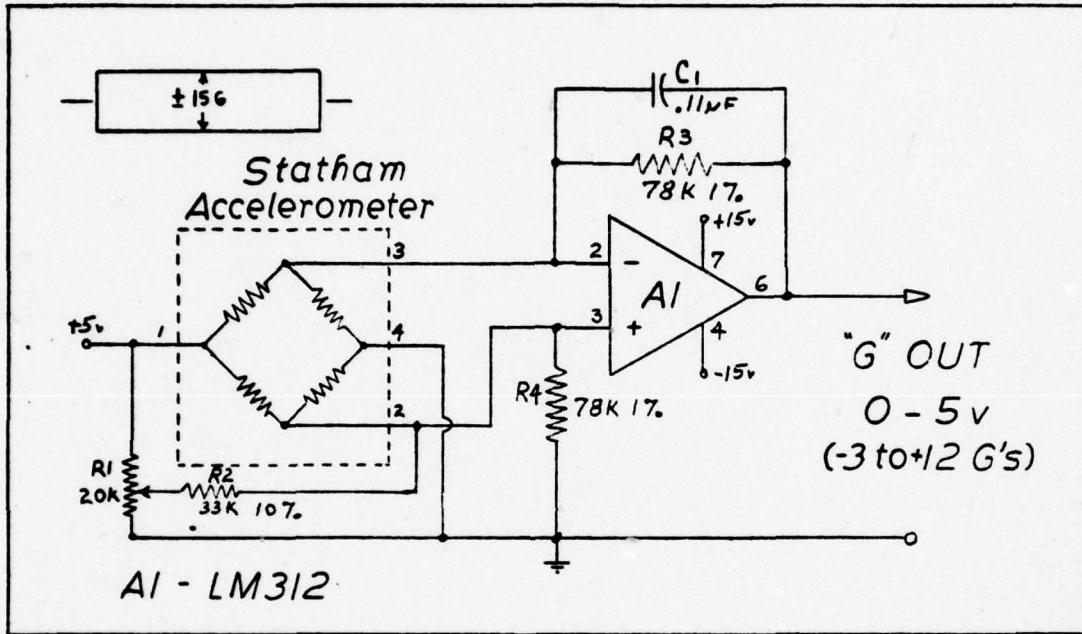


Fig. 23. Accelerometer Interface Circuit

The output range of the circuit is 0 - 5 volts corresponding to -3 to +12 G's. Resistors  $R_1$  and  $R_2$  provide a 1.0 volt (3 G) offset of the bridge. With the accelerometer oriented as shown,  $R_1$  is adjusted to provide a 1.33 volt (1 G) output signal. Feedback network  $C_1$ - $R_3$  determines the amplifier gain; and capacitor  $C_1$  provides filtering to reduce the effects of transient G's. The filter cutoff frequency is approximately 20 Hz.

The output of the amplifier can be tested by changing the orientation of the accelerometer about the longitudinal axis and observing the corresponding signal outputs. With a  $90^\circ$  clockwise rotation, the probe senses a zero-G condition and the amplifier output signal is 1.0 volts. With an additional  $90^\circ$  rotation, the probe senses a negative-one-G condition and the output of the amplifier is .67 volts.

### Heart Rate Detector

The heart rate detector consists of an ECG amplifier, R-wave detector, and digital interval counter. The analog portions are shown in Figure 24; the digital portion is shown in Figure 25.

The ECG amplifier increases a 1 mvolt ECG signal to a minimum of 1.5 volts. The differential ECG (leads 1 and 2) is input to amplifiers  $A_1$  and  $A_2$ , and lead 3 is used as a reference to reduce the common mode voltage. Feedback networks  $R_1-C_1$  and  $R_2-C_2$  provide the gain and high frequency filtering. Potentiometer  $R_4$  allows the gain of the differential input stage to be varied between 30 and 100.

The amplifier signal is capacitively-coupled to amplifier  $A_3$  through  $C_3$  and  $C_4$ . This helps eliminate DC baseline shifts. Amplifier  $A_3$  provides final signal amplification with a differential mode gain of 50. The ECG signal is then filtered through a double-pole, low pass, active filter using amplifier  $A_4$ . The filter bandwidth is 60 Hz and its low frequency gain is unity.

Resistors  $R_1$ ,  $R_2$ , and  $R_7$  to  $R_{10}$  were chosen within .5% tolerance to reduce the common mode amplifier gain. Resistors  $R_5$ ,  $R_6$ ,  $R_{11}$ , and  $R_{15}$  set bias currents for low power (500 microwatts) operation of the MC1776 operational amplifiers.



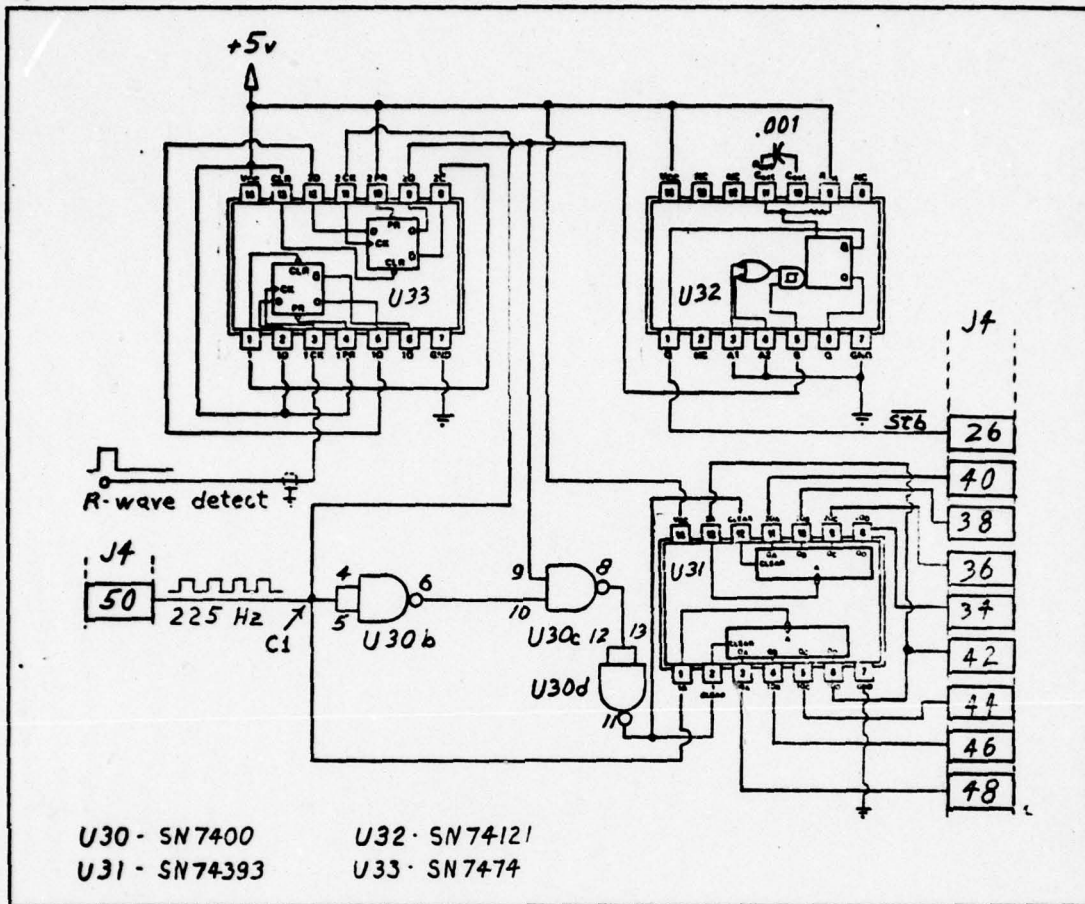


Fig. 25. Digital Interval Counter

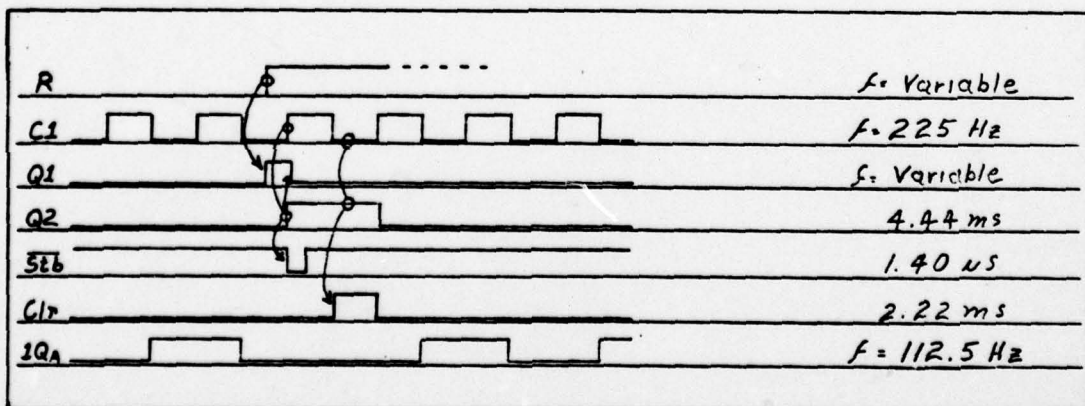


Fig. 26. R-wave Interval Counter Timing Diagram



The filtered ECG signal is input to an inverting comparator ( $A_5$ ). A reference voltage, adjustable by potentiometer  $R_{16}$ , is compared to the ECG input and a pulse is output when the peak of the ECG is above the reference voltage. This pulse is then output to the digital portion of the circuit.

The detector circuit is adjusted by monitoring the ECG on a strip chart or oscilloscope. The amplifier gain is adjusted by  $R_4$  to obtain an ECG peak magnitude of 2 volts. The threshold voltage can be adjusted by  $R_{16}$  to detect all of the R-waves without mis-firing on the noise.

The output of the R-wave detector is input to the digital counting circuitry (Figure 25). This circuit counts the number of 4.44 msec (1/225 Hz) periods between detected R-waves. The dual-D flip-flop configuration (U33) shapes the variable width R-wave pulse to a single 4.44 msec pulse. This pulse is input to a monostable multivibrator (U32) to generate a 1.40 microsec  $\overline{STB}$  for the 8255. The 4.44 msec pulse is also gated to clear the counter prior to subsequent counting pulses. (A timing diagram of this sequence is given in Figure 26.)

The oxygen and accelerometer circuits are biased at  $\pm 15$  volts so that the DAS 1128 power supply can be used. A +5 volt power supply can be used with low power operational amplifiers (i.e., MC1776 or LM312) by splitting the +5 volt power supply to provide the necessary  $\pm 2.5$  volts.

Two OM11 sensor interfaces, the accelerometer interface, and the heart rate detector are contained on a 2.75" x 4.75" component board which is housed in a 3" x 4" x 5" aluminum box. The amplifier and potentiometer orientation is shown in Figure 27. The interface box external connections are shown in Figure 28. The pin assignments for the 25-pin interface connector are given in Table X.

Board #	Part #	Function
Ckt: Heart rate		
1	A1	ECG Gain adj.
2	A2	
3	A3	
4	A4	
5	R4	
6	A5	Threshold adj.
7	R16	
Ckt: PO2IN		
8	A1	Gain adj.
9	R9	
10	R2	Bias adj.
Ckt: PO2CUT		
11	A1	Gain adj.
12	R9	
13	R2	Bias adj.
Ckt: Acceleration		
14	A1	Offset adj.
15	R1	

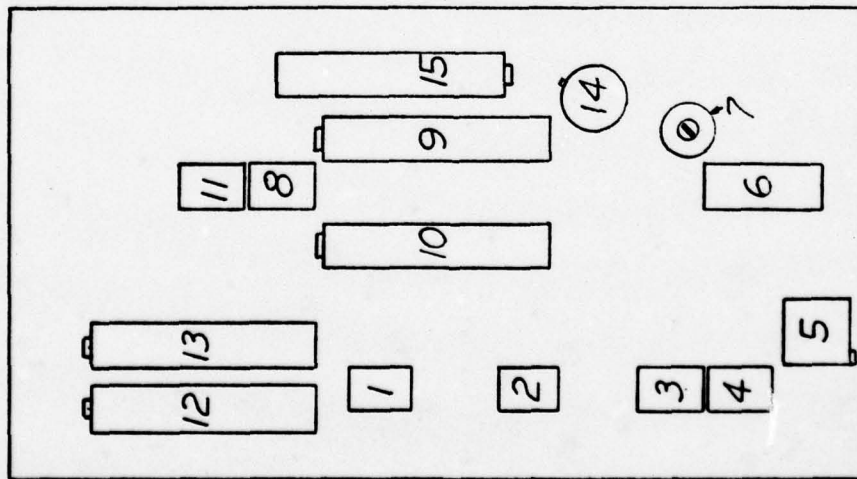


Fig. 27. Chassis Board Layout and Part Number/Function

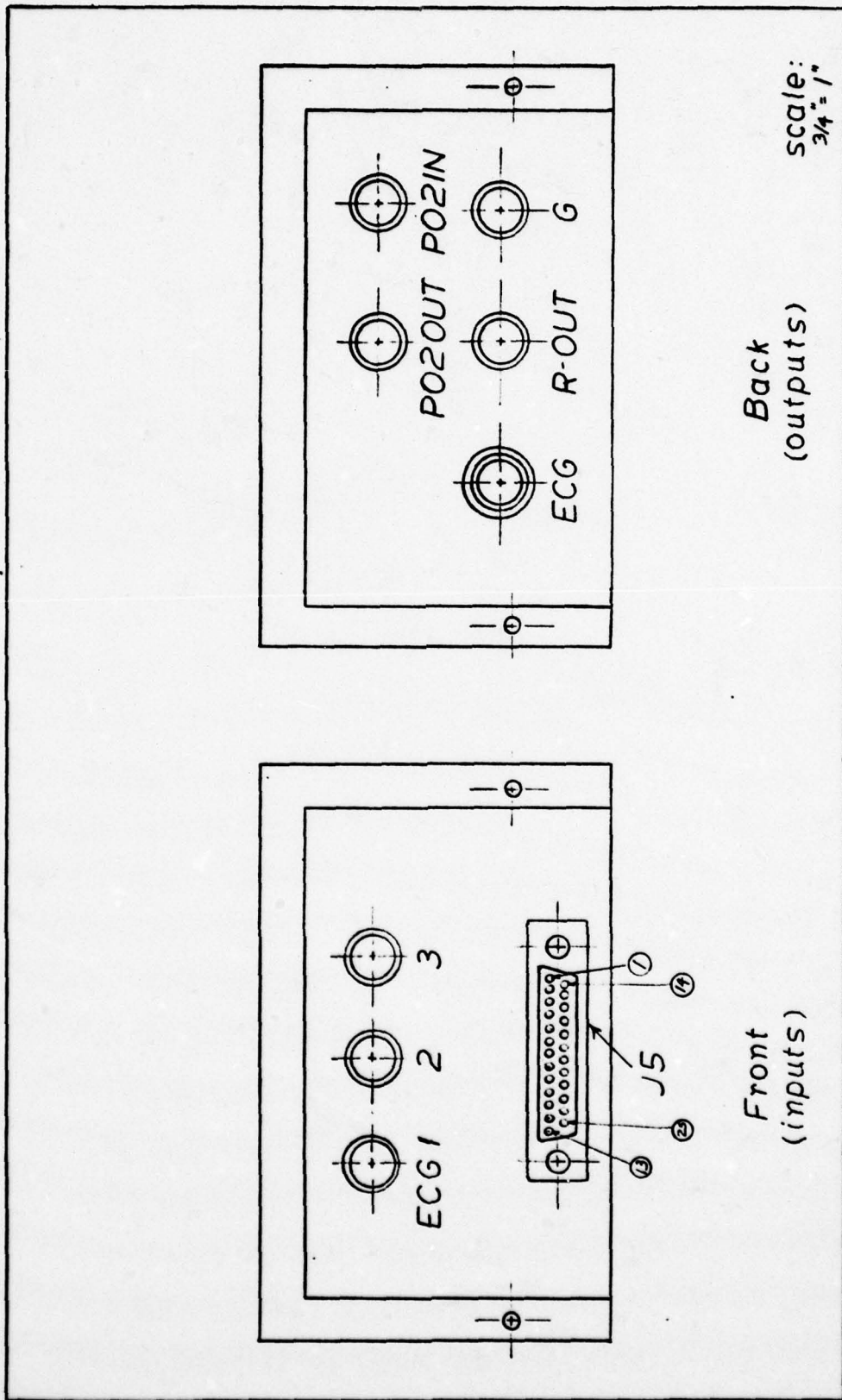


Fig. 28. External Chassis Interface Connections

Table X

Chassis Pin Connections

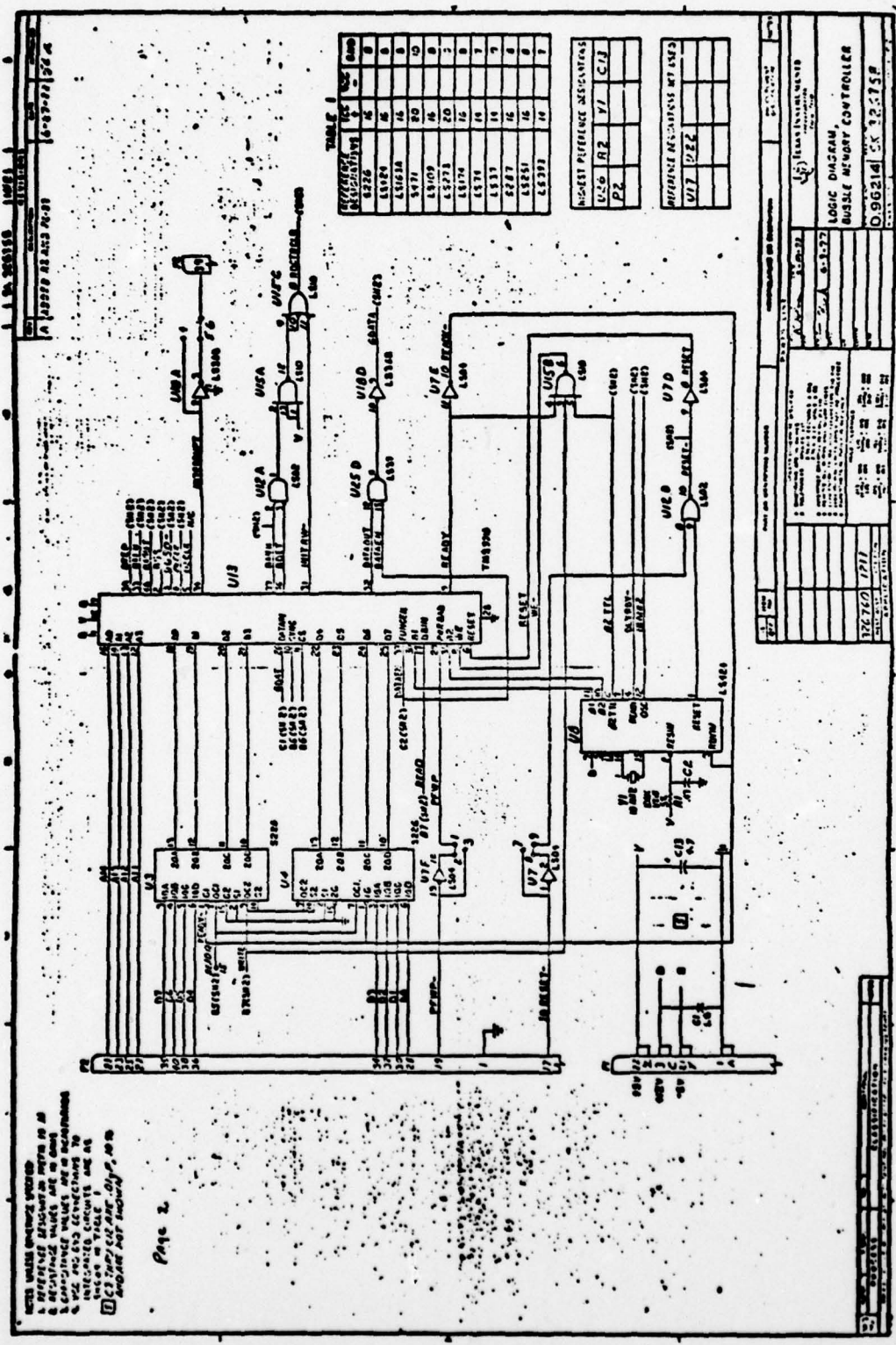
External Connection	Pin #	Chassis Connection
N/C	1	N/C
OM11 #1 (P02IN)	2 3 4 5	} to P02IN amplifier
thermistor (white)		
thermistor (green)		
cathode (red)		
anode (black)		
N/C	6	N/C
N/C	7	N/C
N/C	8	N/C
N/C	9	N/C
accelerometer pin 2	10	} to accelerometer amplifier
accelerometer pin 3	11	
+15 VDC (red)	12	+15 volt bus
-15 VDC (white)	13	-15 volt bus
OM11 #2 (P02OUT)	14 15 16 17	} to P02OUT amplifier
thermistor (white)		
thermistor (green)		
cathode (red)		
anode (black)		
N/C	18	N/C
N/C	19	N/C
OM11 shields	20	GND
N/C	21	GND
accelerometer pin 4	22	GND
GND (black)	23	GND
accelerometer pin 1	24	+5 volt bus
+5 VDC (blue)	25	+5 volt bus

Appendix H

Magnetic Bubble Memory Interface Diagrams

These diagrams show the interfacing required to test the bubble memory with the prototype. The diagrams were supplied by Gerald Cox of Texas Instruments, Inc., Dallas, Texas.

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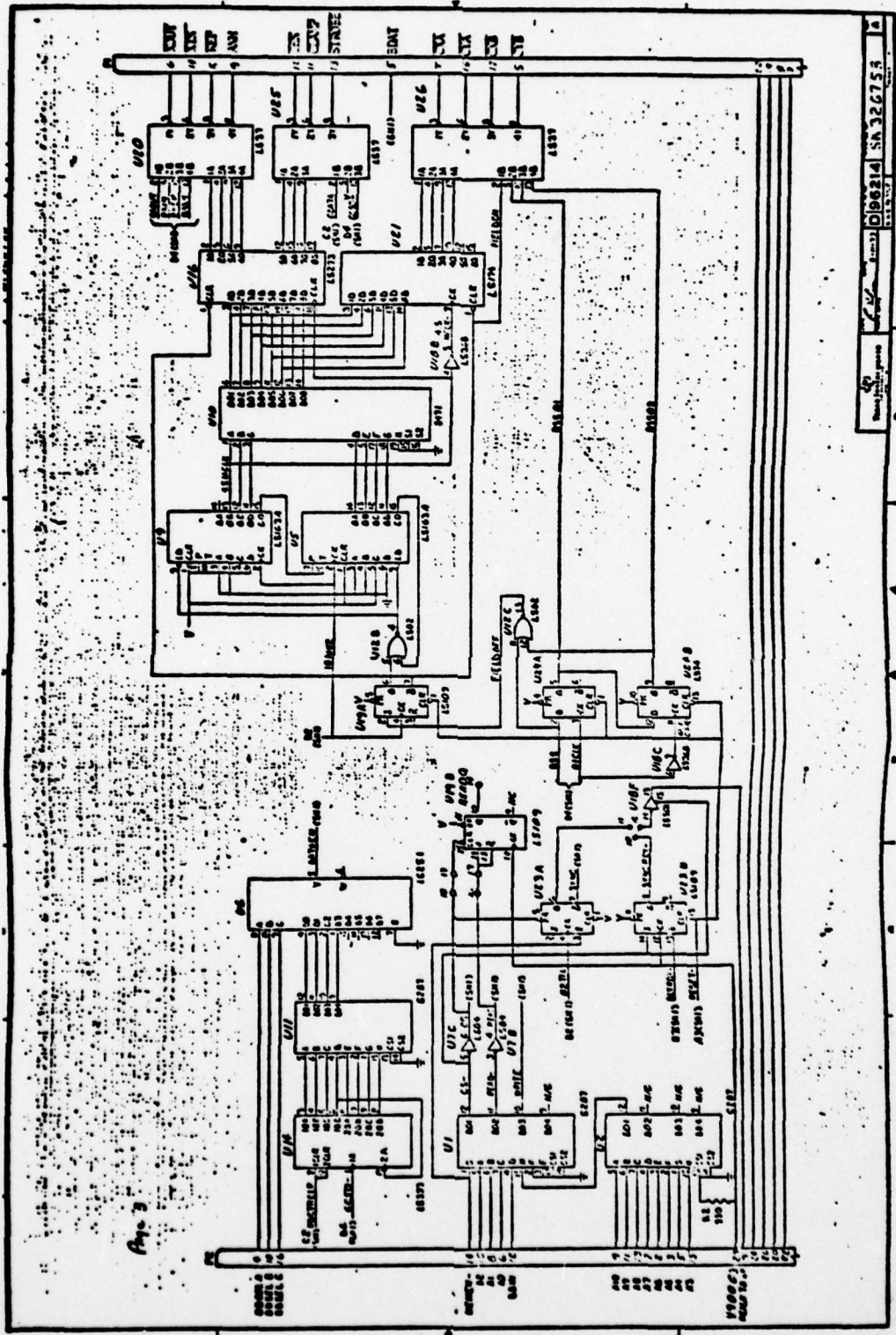


Fig. 30. Logic Diagram, Bubble Memory Controller (Page 2)





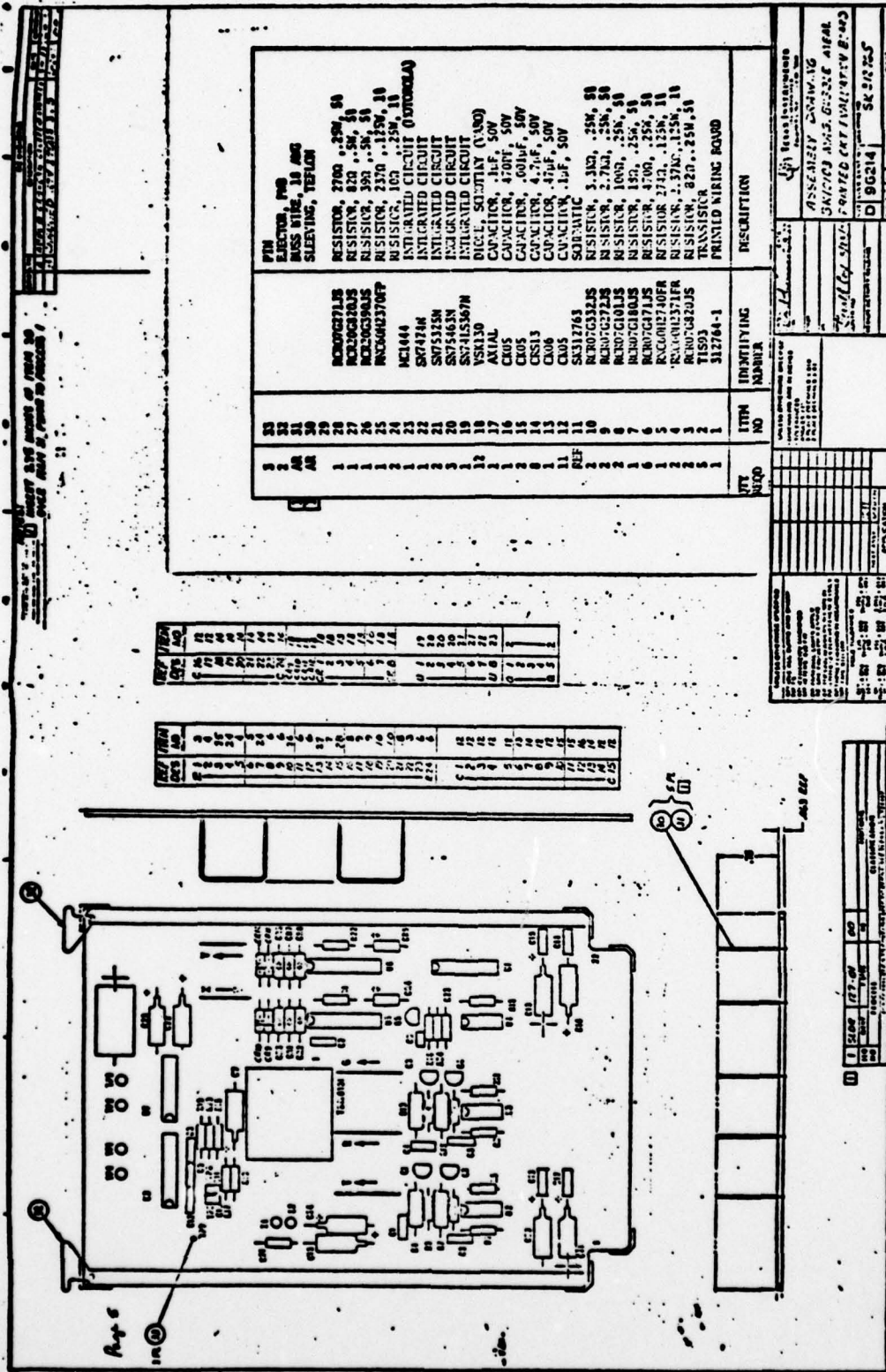


Fig. 32. Magnetic Bubble Memory Printed Circuit Evaluation Board (Page 2)

## Appendix I

### Oxygen Consumption Calculations for Single Breath Analysis

Oxygen consumption is measured by analyzing the mass of the gases of inspiration and expiration. When laboratory conditions permit, precise measurement of inspired and expired flow volumes can be made, and their gaseous contents can be accurately analyzed. Oxygen consumption is then calculated by subtracting the quantity of oxygen exhaled from the quantity of oxygen inhaled (Ref 21:681-685).

Since the IFPDAS II is used in flight, precise analysis of the respiratory gases and oxygen consumption measurements become more difficult. The diluter-demand oxygen regulator provides a mixture of oxygen and air to the pilot. The regulator adds a minimum of 0% oxygen to a maximum of 30% oxygen at sea level. The mixture varies according to outlet flow and altitude as shown in Table XI (Ref 22). Because of varying mixtures, the oxygen consumption analysis involves measuring the inspired and expired volumes and flow-volume averaging the inspired and expired oxygen contents (Ref 23). This method allows a single breath analysis of oxygen consumption using IFPDAS measured parameters. The Beckman OM11 oxygen sensors provide measurements of oxygen partial pressure and the flow sensors provide measurements of flow volume. For this application, the OM11 sensors require the lead compensation discussed in Appendix G which decreases the sensor response time to 100 msec. The remaining discussion presents the mathematical analysis and the IFPDAS II implementation.

Table XI  
Oxygen Ratio

ALTITUDE (1000 FEET)	OUTLET FLOW (LITERS/MINUTE)	% OXYGEN ADDED FROM SOURCE	
		MINIMUM	MAXIMUM
0	15	0	30
0	50	0	30
5	15	1	33
5	50	1	33
10	15	6	45
10	50	6	45
10	135	6	60
15	15	14	52
15	50	14	52
15	135	14	70
20	15	24	55
20	50	24	55
20	135	24	80
25	15	40	80
25	50	40	80
25	135	40	90
28	15, 50, 135	60	100
32	135	98	100
	With diluter at 100%		
All altitudes	15, 50, 135	98	100

(Ref 22)

The instantaneous expired oxygen volume is related to the partial pressure of oxygen by

$$\frac{d V_{E_{O_2}}}{dt} = \frac{d \left\{ V_E \cdot \frac{P_{E_{O_2}}}{P_{Abs}} \right\}}{dt} \quad (22)$$

where,

$V_{E_{O_2}}$  = Volume of expired oxygen (liters)

$V_E$  = Volume of expired air (liters)

$P_{E_{O_2}}$  = Partial pressure of oxygen expired (mm Hg)

$P_{Abs}$  = Absolute pressure (mm Hg)

Since,

$$\frac{P_{E_{O_2}}}{P_{Abs}} = F_{E_{O_2}} \quad (23)$$

where  $F_{E_{O_2}}$  is the oxygen fraction in the expired air.

Using Eq (23), Eq (22) becomes,

$$\frac{d V_{E_{O_2}}}{dt} = \frac{d \left\{ V_E \cdot F_{E_{O_2}} \right\}}{dt} \quad (24)$$

The flow-weighted expired oxygen volume then becomes,

$$V_{E_{O_2}} = \int_{\text{breath}} F_{E_{O_2}} \cdot V_E \, dt \quad (25)$$

where  $V_E$  is the expired volume of oxygen in liters/breath

A similar computation for inspired oxygen volume is made and the per breath oxygen consumption can then be computed:

$$V_{O_2} \text{ (liters/breath)} = V_{I_{O_2}} - V_{E_{O_2}} \quad (26)$$

where,

$$V_{O_2} = \text{Volume of oxygen consumed (liters/breath)}$$

$$V_{I_{O_2}} = \text{Inspired volume of oxygen (liters/breath)}$$

The expired oxygen volume measurement is made using the curves of expired flow volume rate and oxygen partial pressure as shown in Figure 33. The per breath expired oxygen volume is the area under the  $(F_{E_{O_2}})(\dot{V}_E)$  curve measured over the period of one breath. A trapezoidal approximation to the area under this curve will yield the flow-weighted expired oxygen volume:

$$V_{E_{O_2}} = \left[ \frac{1}{2} \left\{ F_{E_{O_2_1}} \cdot \dot{V}_{E_1} \right\} + \sum_{i=2}^{n-1} \left\{ F_{E_{O_2_i}} \cdot \dot{V}_{E_i} \right\} + \frac{1}{2} \left\{ F_{E_{O_2_n}} \cdot \dot{V}_{E_n} \right\} \right] t \quad (27)$$

where,

$$F_{E_{O_2_i}} = \text{Expired oxygen fraction value}$$

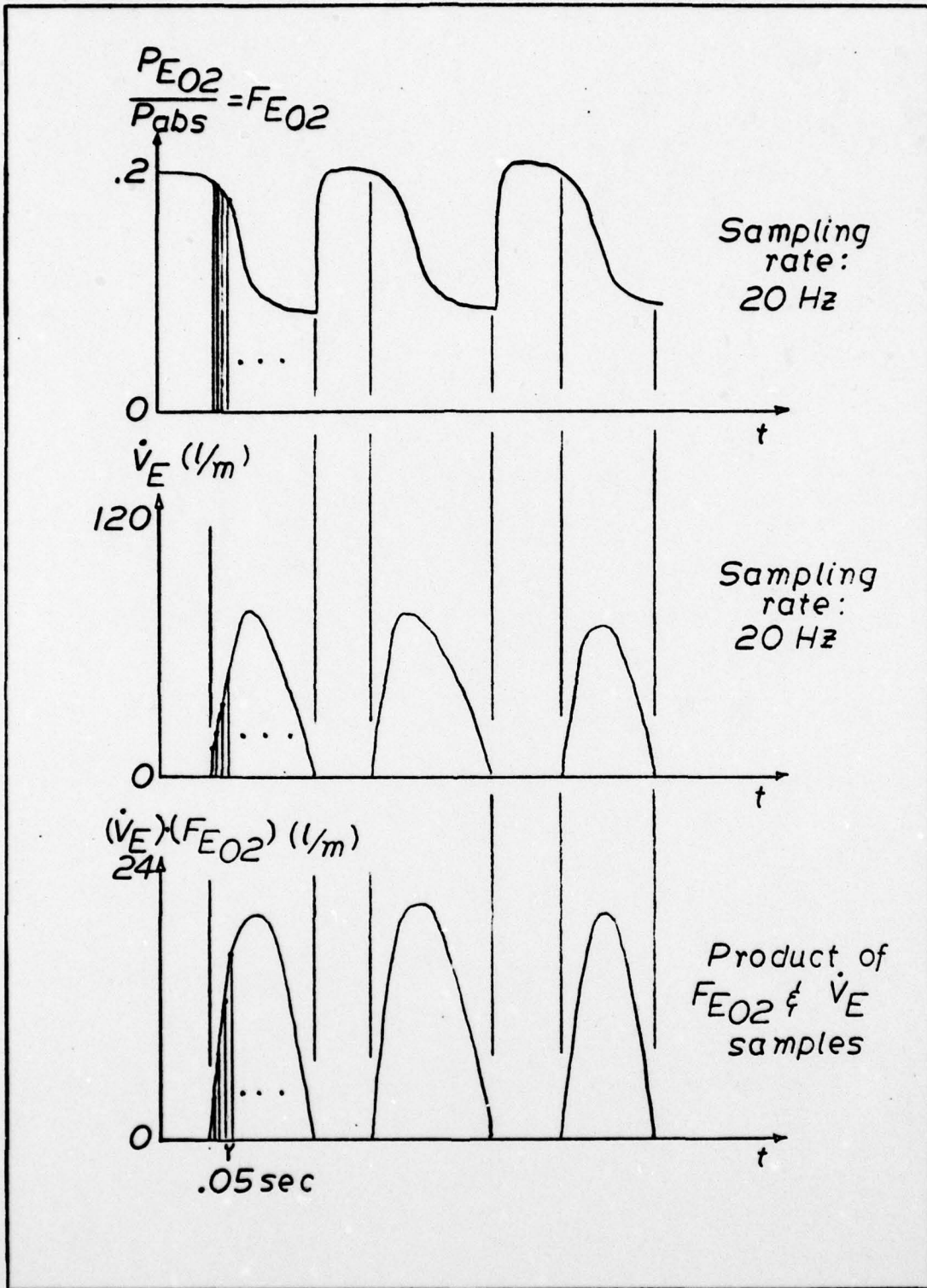


Fig. 33. Flow Rate Weighted Expired Oxygen Measurement

$V_{E_i}$  = Expired volume flow rate sample value (liters/min)

t = Sample to sample width (min)

A similar method is used to determine the inspired oxygen volume and the pilot's oxygen consumption is then computed:

$$V_{O_2} \text{ (liters/min)} = V_{I_{O_2}} - V_{E_{O_2}} \quad (28)$$

The IFPDAS II operating system can be modified to store all of the parameters required for an accurate calculation of the oxygen consumption if an inspired flow sensor is acquired. The required flow-weighted sums are calculated as follows: As the inhaled breath is detected, the inhaled flow rate/oxygen-fraction product is calculated. These samples are summed over the period of the inspired breath and stored. A similar product-sum is calculated for the period of the exhaled breath and stored. These values are then available for altitude correction and integration on the ground.

A sampling rate of 20 Hz is adequate to insure minimal error for the trapezoidal approximation of the integral as the breathing rate averages only .2 Hz. This routine also allows ample time to perform the necessary multiplication and division routines, each requiring a maximum of 380 microsec. Measurement delays can be compensated for by temporarily storing sample values and time-adjusting them to provide maximum accuracy.



## VITA

Joseph Gregory Jolda was born on 12 January 1948 in Worcester, Massachusetts. He graduated from high school in Webster, Massachusetts, in 1965 and attended Northeastern University, Boston, Massachusetts, from which he received the degree of Bachelor of Science in Electrical Engineering in June 1970. Upon graduation he was employed as an associate engineer for Ampere Electronic Corporation, Slatersville, Rhode Island. He also taught Mathematics and Circuit Theory at Worcester Junior College, Worcester, Massachusetts. He then entered active duty in September 1970, and received his USAF commission from OTS in January 1971. He completed pilot training and received his wings in April 1972. He served as a T-33 pilot with the 67th Tactical Reconnaissance Wing at Bergstrom AFB, Texas, and then as a T-37 instructor pilot with the 37th Flying Training Squadron at Columbus AFB, Mississippi, until entering the School of Engineering, Air Force Institute of Technology, in June 1976. He is a member of Eta Kappa Nu and Tau Beta Pi.

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these data are transferred to a land-based computer which completes the data processing and graphs the following environmental and physiological information versus flight time: (1) cabin absolute pressure, (2) cabin altitude, (3) Z-G's, (4) heart rate, (5) breathing rate, (6) minute ventilation volume, (7) inspired oxygen quantity, and (8) expired oxygen quantity.

The completed IFPDAS II prototype provides the desired information well within the required accuracy. It provides the following parameter ranges: (1) heart rate from  $53 \pm .1$  to  $225 \pm 2.2$  b/min, (2) breathing rate from  $4.7 \pm .1$  to  $50 \pm 1$  b/min, (3) minute ventilation volume from 0 to  $100 \pm 2$  l/min, (4) absolute pressure from 0 to  $760 \pm 2$  mm Hg, and (5) G's from -3 to  $+12 \pm .1$  G.

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