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USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK. VOLUME 88. T-33A AIR--ETC(U)  
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Volume 88

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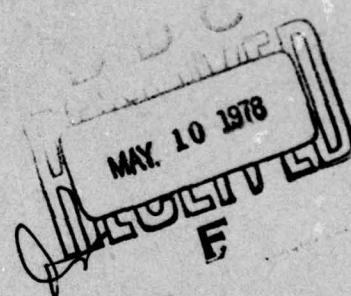
## USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

AD No.  
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Volume 88

T-33A Aircraft, Near and Far-Field Noise

APRIL 1977



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AEROSPACE MEDICAL RESEARCH LABORATORY  
AEROSPACE MEDICAL DIVISION  
AIR FORCE SYSTEMS COMMAND  
WRIGHT-PATTERSON AIR FORCE BASE, OHIO 45433

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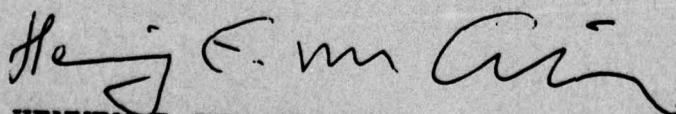
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**HENNING E. VON GIERKE  
Director**

**Biodynamics and Bioengineering Division  
Aerospace Medical Research Laboratory**

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The USAF T-33A is a flight trainer aircraft powered by one J33-A-35 turbojet engine. This report provides measured and extrapolated data defining the bioacoustic environments produced by this aircraft operating on a concrete runup pad for three power conditions. Near-field data are reported for 4 locations in a wide variety of physical and psychoacoustic measures: overall and band sound pressure levels, C-weighted and A-weighted sound levels.			

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preferred speech interference level, perceived noise level, and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Far-field data measured at 19 locations are normalized to standard meteorological conditions and extrapolated from 75-8000 meters to derive sets of equal-value contours for these same seven acoustic measures as functions of angle and distance from the source. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application", AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

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## PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723104, Measurement and Prediction of Noise Environments of Air Force Operations.

The authors gratefully acknowledge Mr. John Cole for his assistance in preparing this report, Mr. Robert England for his assistance in acquiring the raw data, Mr. Keith Kettler, Mr. Henry Mohlman and Mr. David Ellerman of the University of Dayton for assistance in the mechanics of data processing, and Ms. Norma Peachey and Mr. Mike Patterson for assistance in typing and preparation of the graphics.

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## INTRODUCTION

The USAF T-33A is a flight trainer aircraft powered by one J33-A-35 turbojet engine. The aircraft was manufactured by the Lockheed Aircraft Corporation and the engines by Allison, a Division of General Motors Corporation.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the T-33A aircraft.

This volume is one of a series published by the AMRL under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during *ground operations* of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15 C temperature, 70% rel humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. Refer to Volumes 1 and 2 (references 1 and 2) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

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1. Cole, John N., *USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application*, AMRL-TR-75-50 (1) Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.
  2. Cole, John N., *USAF Bioenvironmental Noise Data Handbook, Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise*, AMRL-TR-75-50 (2), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

## NEAR-FIELD NOISE

### MEASUREMENTS

AMRL acquired near-field noise data on the T-33A aircraft during ground runup operations of its turbojet engine. For these tests the aircraft was located on the "Hot Cargo" pad, Eglin AFB, FL, with no significant reflecting surfaces in the vicinity except the ground plane. Table 1 gives the surface meteorological conditions and the four engines and ground support equipment power conditions. The ground-crew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand-held microphone in and around each location, probing all areas where a crew member's head would normally be located. He recorded all the noise samples on magnetic tape. During analysis of each sample he determined the one-third octave band root-mean-square sound pressure using a 4- or 8-second integration time to derive a power-averaged level for each location. Figure 1 shows the four near-field locations where ground crew are usually located for maintenance and/or preflight checkout operations. Estimates of noise levels at other locations are difficult in the near-field since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A means ground crew location 1 and test condition A.

### RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the T-33A aircraft at the four ground crew locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data one can calculate the variety of measures given in Table 3, which are widely used to assess the effects of noise on personnel and their performance.

All near-field data are for the meteorological conditions at the time of test but are valid for all typical airbase meteorology because of the short sound propagation distances involved.

TABLE 1  
MEASUREMENT LOCATIONS AND TEST CONDITIONS  
FOR NEAR-FIELD NOISE MEASUREMENTS

T-33A Aircraft, Ground Runup, Eglin AFB  
Tail # 63655, 15 July 1971

*Ground Crew Location*

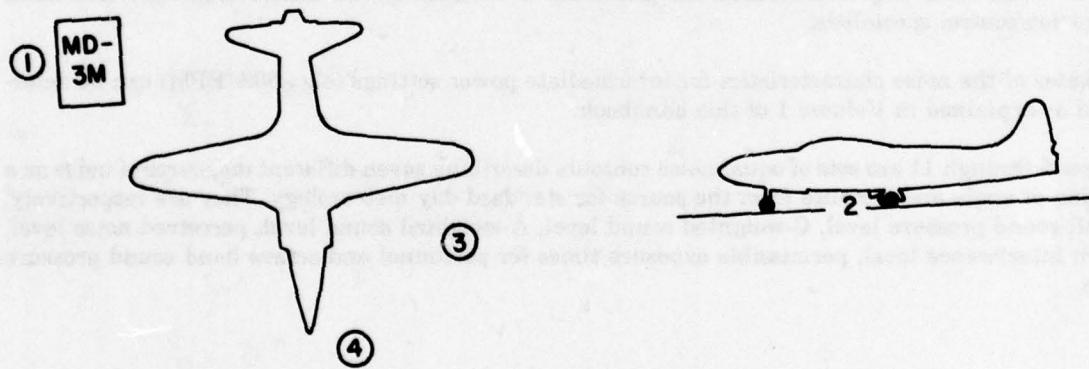
1	Operator MD-3M
2	Wheel Chock Pull
3	Marshall
4	Crew Chief Observer

*Aircraft Engine (and Support Equipment) Operation*

A	MD-3M Operating (unloaded)
B	Engine Start, MD-3M Operating (loaded)
C	Idle Power and MD-3M (unloaded)
D	Idle Power

*Meteorology*

Temperature	25.6 C
Bar Pressure	0.758 M Hg
Rel Humidity	85 %
Wind — Speed	1 M/Sec (2 Kt)
— Direction	270 Deg.



**Figure 1. Near-Field Measurement Locations on the Hot Cargo Pad at Eglin AFB FL**

## FAR-FIELD NOISE

### MEASUREMENTS

AMRL acquired the near and far-field data during a 1- 2-hour test period, thus keeping similar meteorological conditions. Figure 2 shows the aircraft on the "Hot Cargo" pad and its orientation relative to 19 microphone measurement sites on a semicircle. The center of the 75 meter radius semicircle used in surveying the J33-A-35 engine was on the ground directly below the intersection of the aircraft's centerline and the plane passing through the engine's exhaust plane. The ground runup pad did not have a blast deflector; therefore, the jet exhaust was in a "free-flow" condition.

Table 4 provides cockpit readouts of engine speed in percent for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All 19 microphone measurement sites are in the acoustic far-field of their respective source where the sound wave-fronts spherically diverge and the noise source may be regarded as a point source.

A portable microphone/tape recorder system was used to sequentially record 5 to 10 seconds of noise at each far-field location. The microphone was hand-held 1.7 meters (5-1/2 feet) above the ground and pointed at the source ( $0^\circ$  angle of incidence). These samples were then time-integrated to derive a root-mean square sound pressure level.

### RESULTS

Table 5 lists the overall and 1/3 octave band SPL measured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 100 meters distance and standard meteorological conditions (15 C temperature, 70% relative humidity, 0.760 meter Hg barometric pressure) and used to derive the graphic data in Figure 3 which provides a compact summary of the far-field noise characteristics of the T-33A aircraft in a standard format.

Figure 4 and Table 6 present two basic acoustic measures, the acoustic power levels and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure that describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

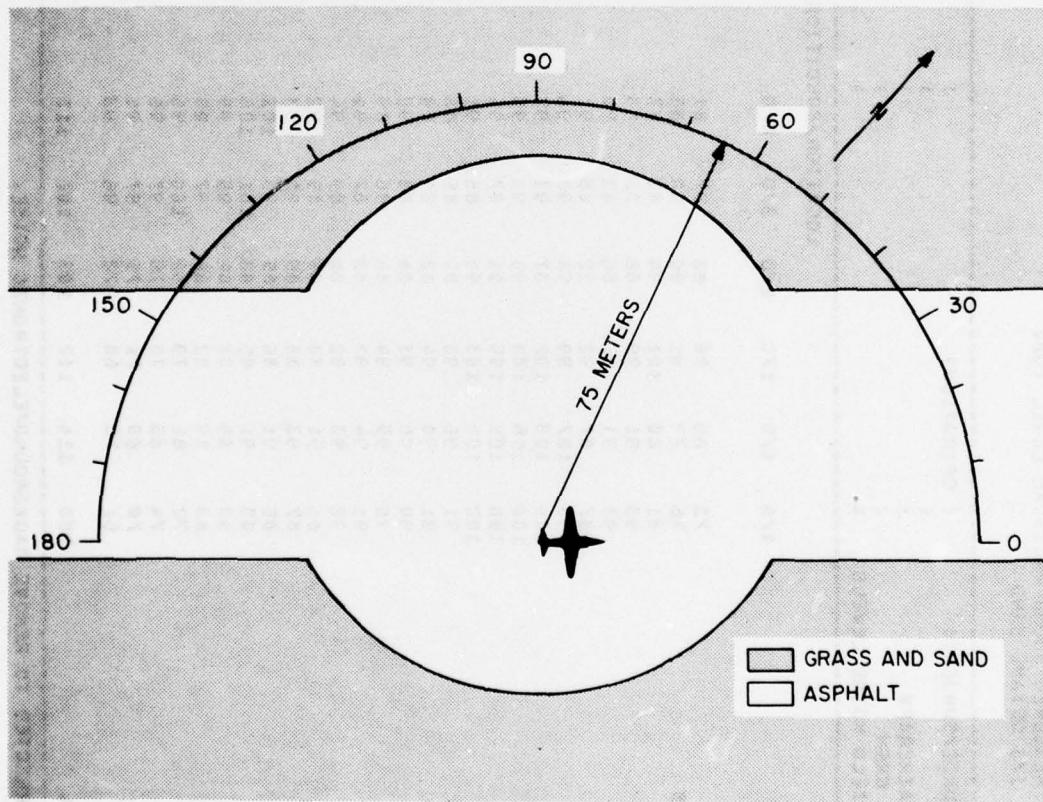
Estimates of the noise characteristics for intermediate power settings (e.g., 80% RPM) can be determined as explained in Volume 1 of this handbook.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

Data excessively influenced by spurious background/electronic noise were eliminated from all figures and tables. No data are presented beyond the 160 degree location for the idle power settings because of background/electronic noise. Typically, the A-weighted levels for these angles are from 5 to 10 dBA below the level at the 160 degree location.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating. Data eliminated because they were near the background/electronic noise were generally not significant because the levels were so low.

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.



**Figure 2. Far-Field Measurement Locations on the Hot Cargo Pad at Eglin AFB, FL**

TABLE I  
1/3 OCTAVE BAND  
2

T-33A AIRCRAFT  
GROUND CREW  
NEAR FIELD NOISE LEVELS

NOISE SOURCE/SUBJECT:	OPERATION:	LOCATION/CONDITION					
FREQ (HZ)		1/A	1/B	1/C	2/D	3/D	4/D
25.5		73	69	85	83	77	81
40.0		76	72	91	90	80	85
50.0		81	80	101	94	82	87
63.0		93	91	99	88	82	88
80.0		87	87	97	86	83	90
100.0		96	107	92	92	85	91
125.0		100	108	99	93	93	96
160.0		100	106	102	87	91	97
200.0		100	105	105	91	87	91
250.0		102	106	103	90	86	91
315.0		91	96	93	86	85	88
400.0		91	98	94	82	86	89
500.0		90	96	93	84	88	90
630.0		90	95	94	86	86	89
800.0		91	94	92	83	87	91
1000.0		92	93	92	88	89	94
1250.0		86	91	88	95	89	92
1600.0		87	93	88	85	90	93
2000.0		85	91	86	85	95	100
2500.0		83	91	85	88	101	106
3150.0		80	89	82	86	95	95
4000.0		80	89	82	86	97	95
5000.0		77	86	79	82	100	96
6300.0		74	83	76	79	97	96
8000.0		70	80	71	75	97	94
10000.0		67	75	68	72	96	93
OVERALL		108	114	112	103	107	110

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE: MEASURED SOUND PRESSURE LEVEL (dB)

2 OCTAVE BAND

NOISE SOURCE/SUBJECT:	( OPERATION:	LOCATION/CONDITION					
		1/A	1/B	1/C	2/B	3/D	4/F/D
T-33A AIRCRAFT							
GROUND CREW							
NEAR FIELD NOISE LEVELS							
FREQ (Hz)							
31.5	63	83	81	101	95	85	89
63	96	95	102	94	88	94	
125	104	112	109	95	96	100	
250	104	109	107	94	91	95	
500	95	101	98	89	91	94	
1000	95	97	96	96	93	97	
2000	90	96	91	91	102	107	
4000	84	93	86	89	102	100	
8000	76	85	77	81	101	99	
OVERALL		106	114	112	103	107	110

IDENTIFICATION:	OMEGA 3.2
TEST	71-019-106
RUN	81
04 DEC 74	
PAGE	J1

TABLE I MEASURES OF HUMAN NOISE EXPOSURE

3

		IDENTIFICATION			
NOISE SOURCE/SUBJECT	OPERATION	1	2	3	4
T-33A AIRCRAFT		OMEGA 3.2			
GROUND CREW		TEST 71-019-100			
NEAR FIELD NOISE LEVELS		RUN 01			
		04 DEC 74			
		PAGE H1			
LOCATION/CONDITION					
	1/A	1/B	1/C	2/D	3/D
HAZARD/PROTECTION					
C-WEIGHTED OVERALL SOUND LEVEL (OASLC IN DB) AT EAR					
A-WEIGHTED OVERALL SOUND LEVEL (OASLA IN DB) AT EAR					
MAXIMUM PERMISSIBLE TIME (T IN MINUTES) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)					
NO PROTECTION					
OASLC	106	114	112	102	106
OASLA	100	105	103	99	107
T	30	13	15	36	9
MINIMUM OPL EAR MUFFS					
OASLA*	85	92	90	78	82
T	404	120	170	960	679
AMERICAN OPTICAL 1700 EAR MUFFS					
OASLC*	81	87	65	73	76
T	887	285	404	960	960
V-51R EAR PLUGS					
OASLC*	78	63	81	73	77
T	960	571	687	960	960
AMERICAN OPTICAL 1700 EAR MUFFS PLUS V-51R EAR PLUGS					
OASLC*	64	69	68	61	65
T	960	960	960	960	960
H-133 GROUND COMMUNICATION UNIT					
OASLC*	75	81	79	73	80
T	960	887	960	960	460
COMMUNICATION PREFERRED SPEECH INTERFERENCE LEVEL (PSIL IN DB)					
PSIL	93	96	95	92	95
ANNOYANCE PERCEIVED NOISE LEVEL, TONE CORRECTED (PNLT IN PNDB)					
TOKE CORRECTION (C IN DB)					
PNLT	115	121	117	115	123
C	1	1	1	3	2
					3

\* BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE.

TABLE 4  
TEST CONDITIONS  
FOR FAR-FIELD NOISE MEASUREMENTS

T-33A Aircraft Ground Runups, Eglin AFB, FL  
Tail # 63655, 15 July 1971

*Aircraft Engine Operation*

Idle Power	35 % RPM
Runup Power	50 % RPM
Military Power ↓	100 % RPM

*Meteorology*

Temperature	25.6 C
Bar Pressure	0.758 M Hg
Rel Humidity	85 %
Wind — Speed	1 M/Sec (2 Kts)
— Direction	270 Deg



TABLE I  
MEASURED SOUND PRESSURE LEVEL (DB)  
5 1/3 OCTAVE BAND DISTANCE = 75 METERS

NOISE SOURCE/SUBJECT:			OPERATION:			METEOROLOGY:			TEST 75-002-045		
	T-33A AIRCRAFT	J33-A-35 ENGINE FAR FIELD NOISE	( 50% RPM	( FREE FLOW	(	) TEMP = 26 C	) BAR PRESS = .758 MM HG	) REL HUMID = 85 %	) 09 MAY 75	) PAGE 2	
FREQ (HZ)	0	10	20	30	40	50	60	70	80	90	100
25	66<	68<	67<	66<	66<	67<	68<	70<	71<	74<	75<
31.5	68<	73<	71<	68<	69<	70<	71<	72<	72<	75<	79
40	67<	75<	74<	71<	73<	74<	72<	73<	75<	76	80
50	68<	71<	73	70<	71<	73<	72<	73	74	76	82
63	71<	72<	71<	71<	73<	74<	73<	74<	76<	80	84
80	75	76	74	74	75	77	76	78	79	82	84
100	63	79	79	79	79	80	80	82	83	85	87
125	79	79	78	79	78	79	79	81	83	84	85
160	78	77	78	79	76	76	78	79	80	81	83
200	77	75	76	75	75	76	75	76	79	81	82
250	75	75	75	76	77	76	77	78	79	80	83
315	74	75	76	77	77	77	77	79	81	84	86
400	74	75	76	77	77	76	76	80	80	82	85
500	74	75	77	78	79	77	79	80	81	82	84
630	71	74	75	77	79	76	80	80	82	83	85
800	72	76	77	78	78	79	82	83	85	85	88
1000	69	72	72	74	75	74	77	80	82	86	87
1250	71	72	72	73	74	73	76	78	80	86	87
1600	72	71	70	73	73	75	74	77	78	82	85
2000	75	71	69	71	73	74	74	77	82	87	91
2500	80	76	74	76	77	74	72	70	74	81	84
3150	74	70	70	73	72	71	71	70	72	76	79
4000	72	71	72	74	72	70	71	69	70	79	80
5000	76	74	75	75	74	71	73	71	70	73	78
6300	71	70	70	71	68	70	69	67	69	76	80
8000	70	69	68	69	70	67	69	68	73	84	87
10000	65	64	64	66	64	65	63	60	62	70	71
OVERALL	89	88	88	89	89	89	90	91	93	94	97

< LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE I MEASURED SOUND PRESSURE LEVEL (DB)  
5 1/3 OCTAVE BAND DISTANCE = 75 METERS

NOISE SOURCE/SUBJECT:					OPERATION:				METEOROLOGY:								
									TEMP = 26 C	BAR PRESS = .756 H HG	REL HUMID = 85 %						
FREQ (HZ)	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	
T-33A AIRCRAFT	70<	73<	71<	72<	75<	74<	77<	78	79	82	83	85	89	92	93	91	78
J33-A-35 ENGINE	73<	73<	75<	74<	75<	75<	76	78	79	81	83	84	86	92	93	91	79
FAR FIELD NOISE	75<	75<	76	75<	78	79	80	80	81	82	85	86	94	96	96	93	76<
50000 63000 100000	96	97	98	99	100	101	102	103	104	105	106	107	108	109	110	111	112

< LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

IDENTIFICATION:			
TEST 75-002-045		OMEGA 1.4	
RUN 03		RUN 03	
09 MAY 75		PAGE 2	

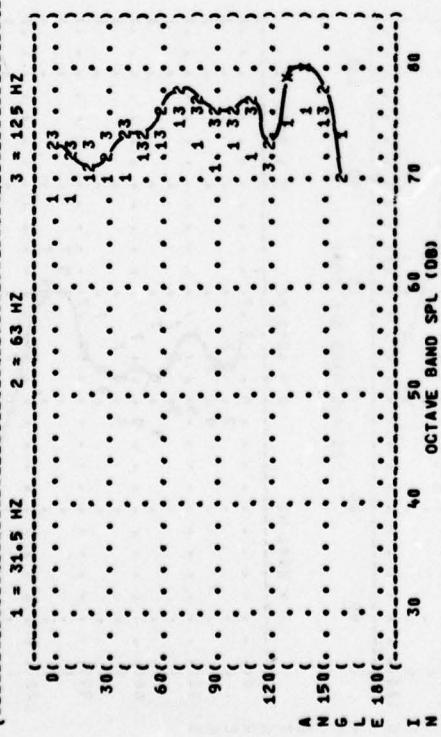
FIGURE 1 NORMALIZED FARFIELD NOISE LEVELS

3 DISTANCE = 100 METERS

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

1 = 31.5 Hz      2 = 63 Hz      3 = 125 Hz



25

IDENTIFICATION

OMEGA 1.4  
TEST 75-012-045  
RUN 01

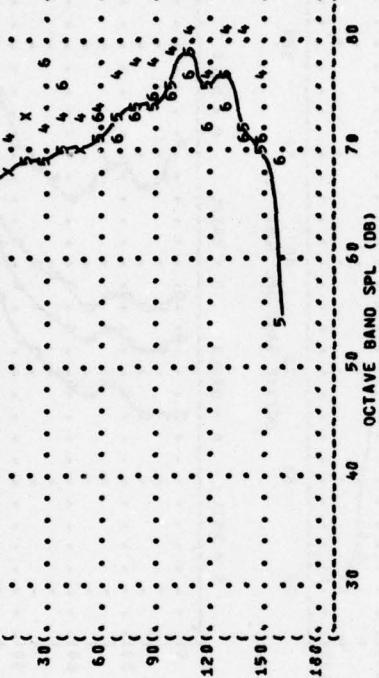
OPERATION:

100% POWER  
35X RPM  
FREE FLOW

METEOROLOGY:

TEMP = 15 C  
BAR PRESS = .760 MM HG  
REL HUMID = 70 %

6 = 1000 Hz

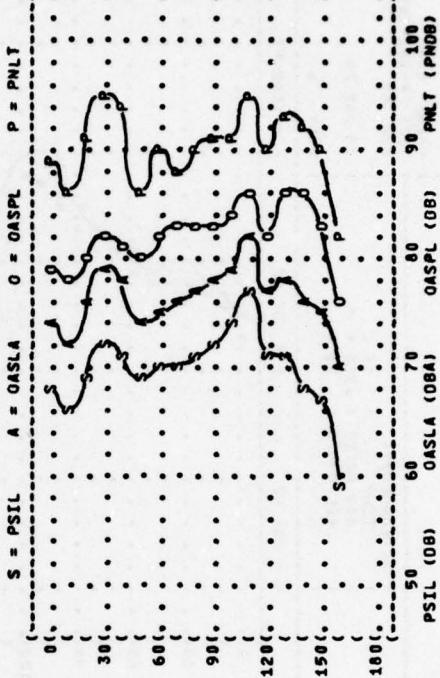


4 = 250 Hz

5 = 500 Hz

6 = 1000 Hz

OCTAVE BAND SPL (DB)

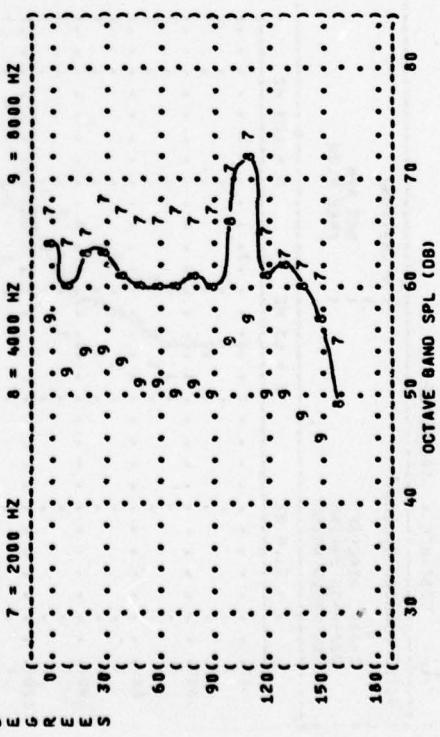


OCTAVE BAND SPL (DB)

PSIL (OB)

OASPL (OB)

PNLT (PNDdB)



OCTAVE BAND SPL (DB)

PSIL (OB)

OASPL (OB)

PNLT (PNDdB)

FIGURE 1 NORMALIZED FARFIELD NOISE LEVELS

3 DISTANCE = 100 METERS

NOISE SOURCE/SUBJECT: T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION: 50% RPM  
FREE FLOW

1 = 31.5 Hz 2 = 63 Hz 3 = 125 Hz 4 = 250 Hz 5 = 500 Hz 6 = 1000 Hz

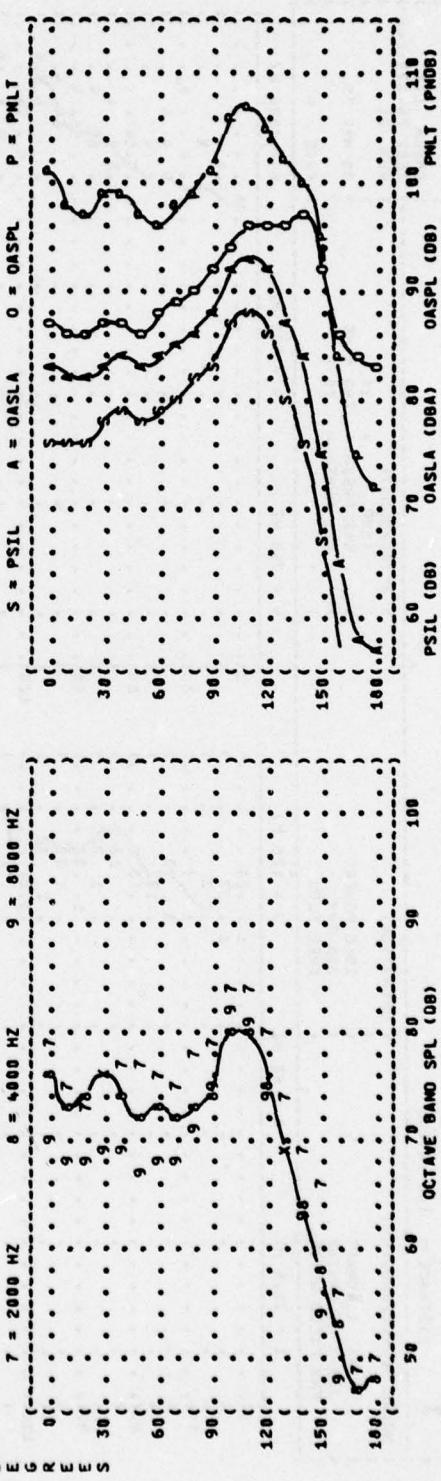
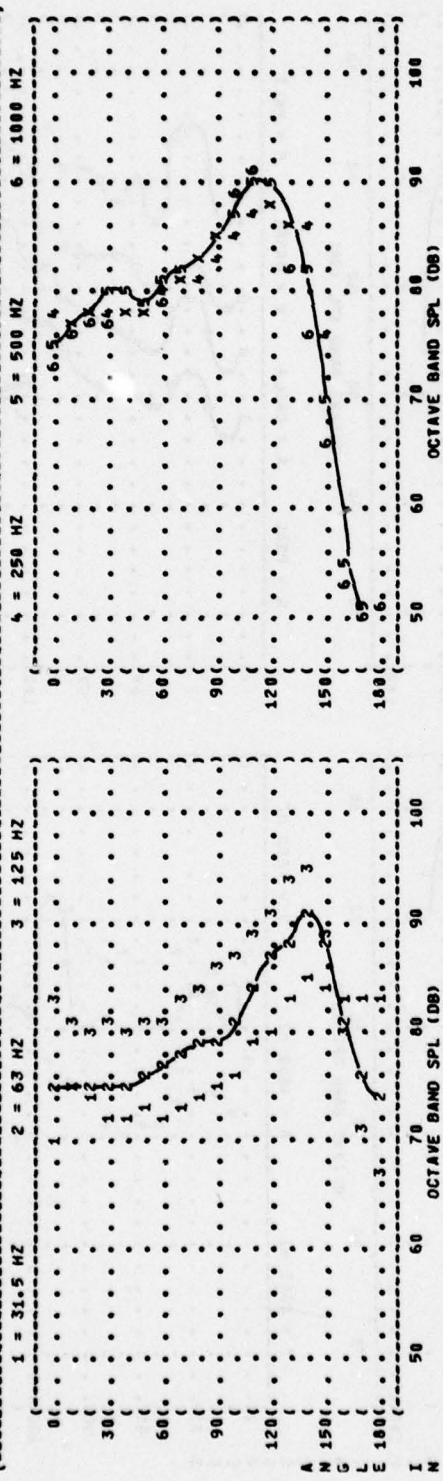


FIGURE: NORMALIZED FARFIELD NOISE LEVELS

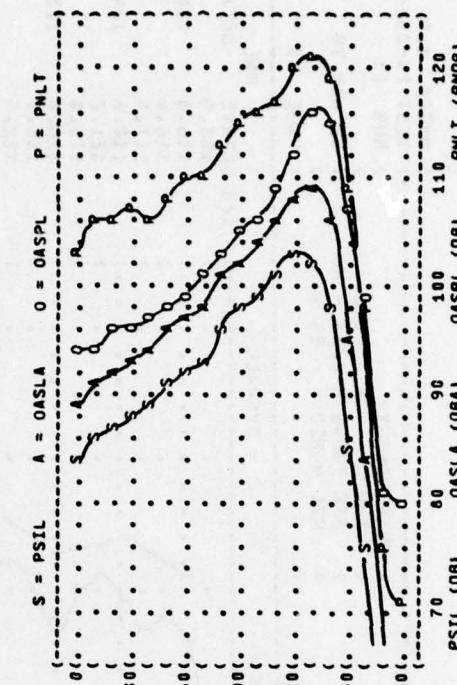
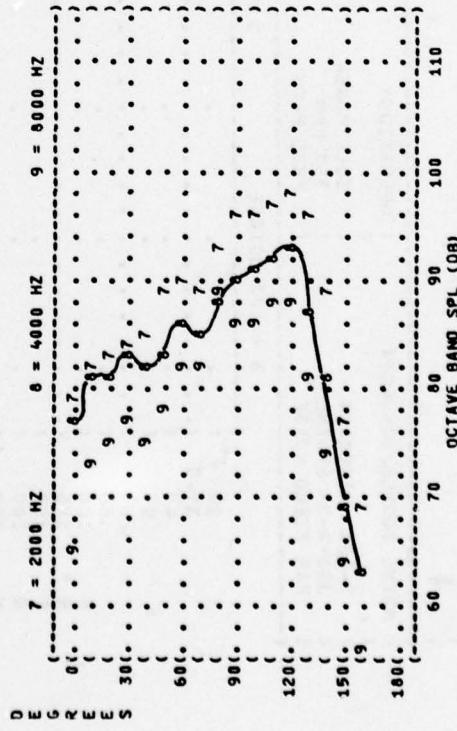
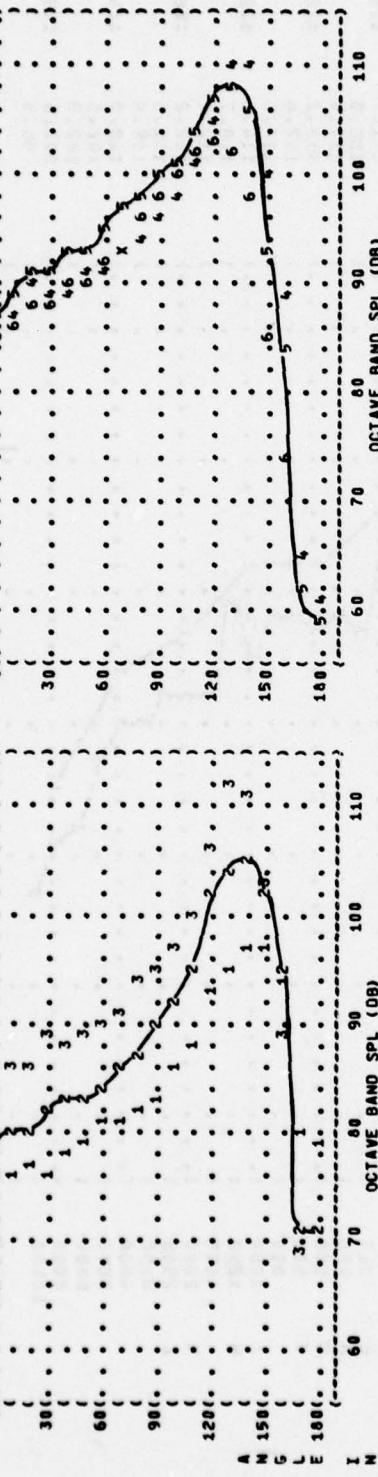
3 DISTANCE = 100 METERS

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

1 = 31.5 Hz      2 = 63 Hz      3 = 125 Hz

4 = 250 Hz      5 = 500 Hz      6 = 1000 Hz



IDENTIFICATION:

OMEGA 1.4

TEST 75-002-045

RUN 03

09 MAY 75

PAGE 6

OPERATIONS:

MILITARY POWER

100% RPM

FREE FLOW

1 = 31.5 Hz      2 = 63 Hz      3 = 125 Hz

4 = 250 Hz      5 = 500 Hz      6 = 1000 Hz

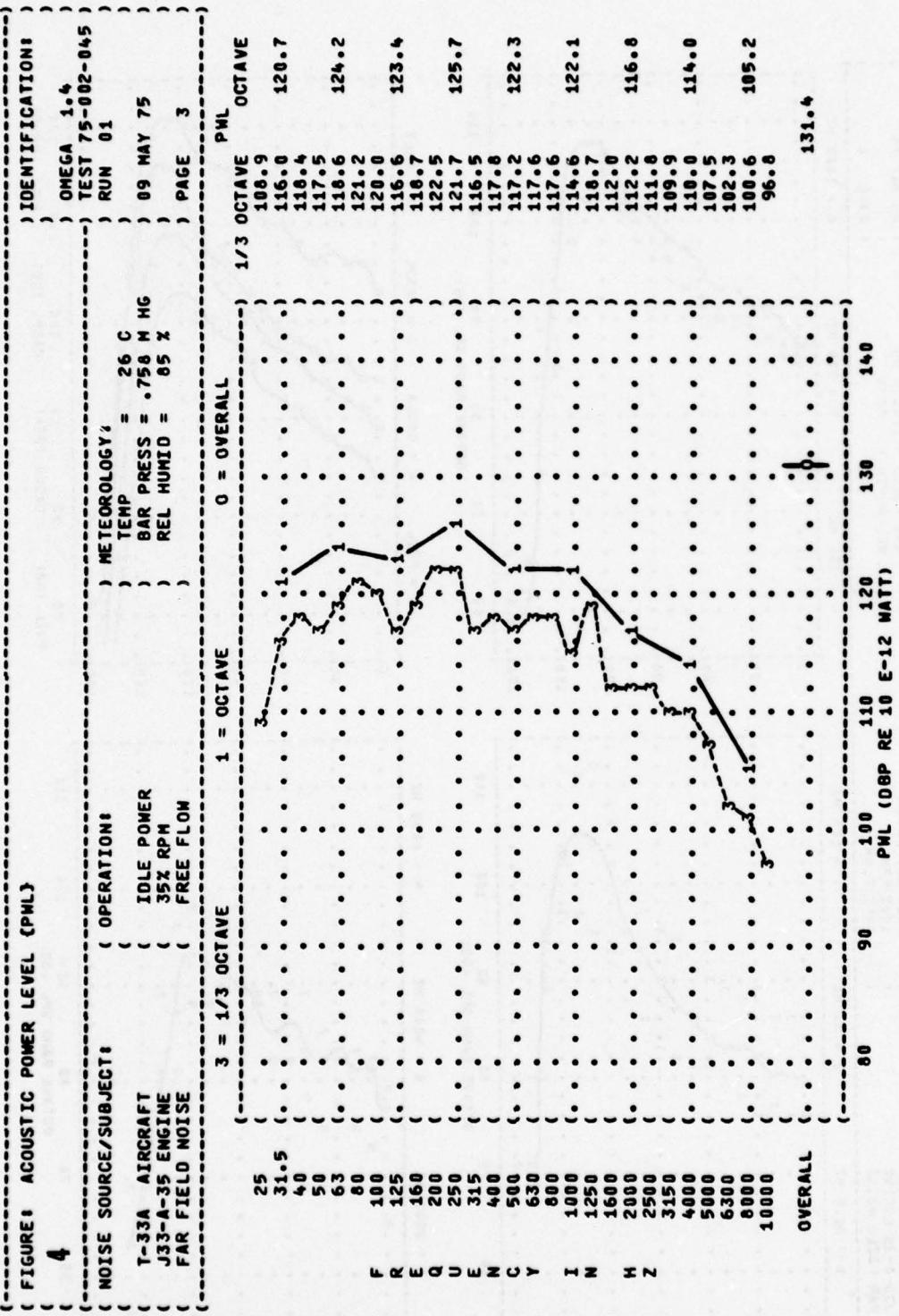
METEOROLOGY:

TEMP = 15 C

BAR PRESS = .760 Hg

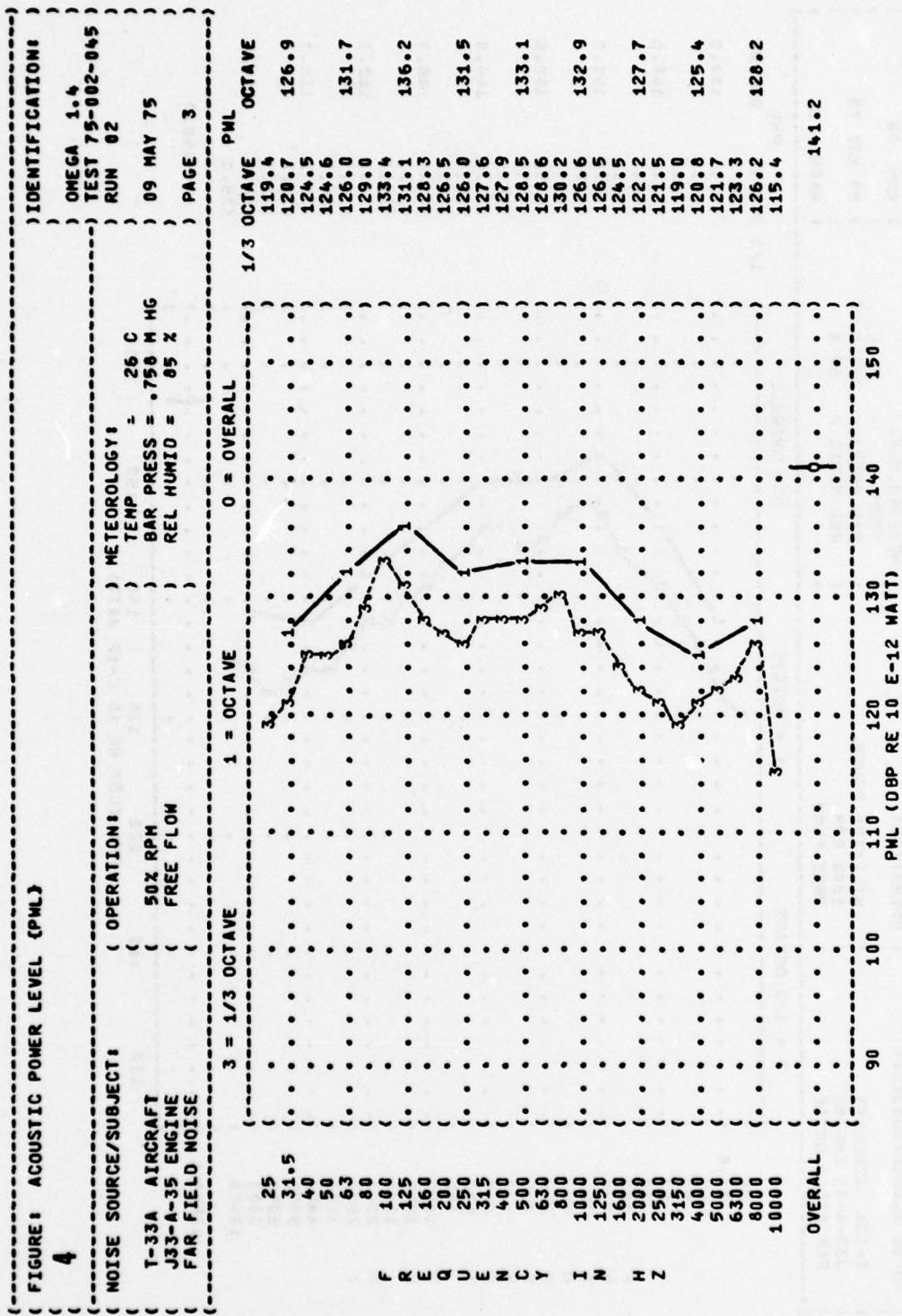
REL HUMID = 70%

{ FIGURE 4 ACOUSTIC POWER LEVEL (PWL)



{ FIGURE: ACOUSTIC POWER LEVEL (PWL) }

4



{ FIGURE: ACOUSTIC POWER LEVEL (PWL)

4

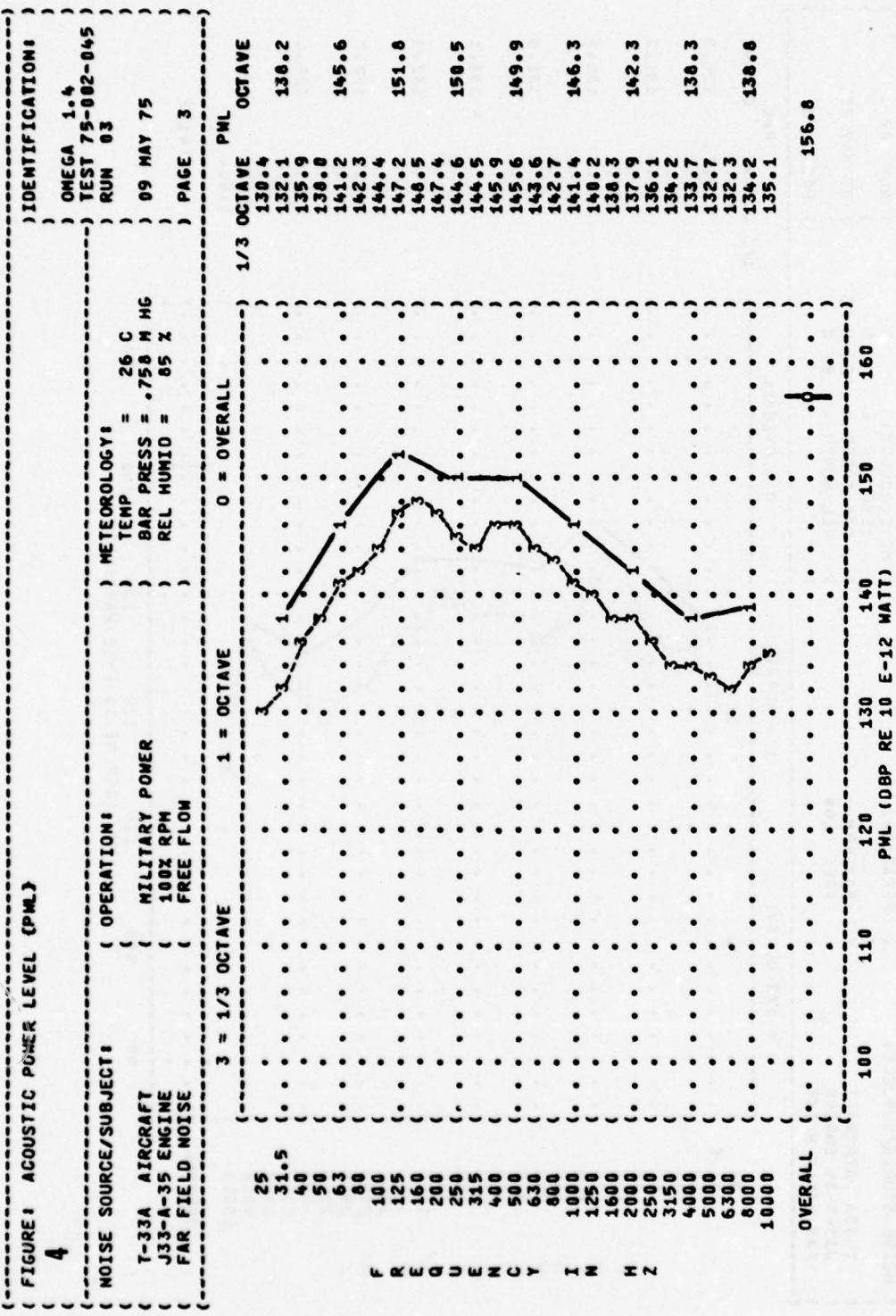


TABLE I DIRECTIVITY INDEX (DB)

6

IDENTIFICATION:			
OMEGA 1A TEST 75-002-045			
RUN 01			
NOISE SOURCE/SUBJECT:	OPERATION:	METEOROLOGY:	
T-33A AIRCRAFT J33-A-35 ENGINE FAR FIELD NOISE	IDLE POWER 35% RPM FREE FLOW	TEMP = 26 C BAR PRESS = .758 Hg REL HUMID = 85%	
		PAGE 4	
FREQ (HZ)	0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180	ANGLE (DEGREES)	
1/3 OCTAVE	0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180	0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180	
25	3 -2 -2 0	3 0 -2 1 -1 0 1 2 3 4 4 5 4 4 2 1	
31.5	3 -4 -4 -2 -2 0	3 0 -1 1 -2 1 1 1 1 1 2 4 3 4 3 0	
40	-5 -6 -5 -6 -7 -7 -6 -5 -7 -6 -3 -3 -2 -2 -3 -4	-2 -1 0 1 -1 0 1 1 1 0 2 4 3 5 2 -6	
50	-7 -7 -6 -6 -5 -5 -3 -3 -2 -3 -1 -1 0 -1 -1 -1	-1 0 1 1 0 1 1 1 1 0 1 1 3 4 2 -7	
63	-5 -5 -3 -3 -3 -2 -3 -3 -2 -2 -2 -1 -1 0 -1 -1	-1 0 1 1 1 1 1 1 1 1 1 1 3 3 3 1	
80	-2 -3 -2 -3 -2 -2 -1 -2 -1 -1 -2 0 1 0 -1 1	0 -1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
100	-1 -2 -2 -1 -2 -1 -1 -2 -1 -1 -2 0 1 0 -1 1	0 -1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
125	-1 -3 -2 -3 -2 -3 -2 -1 -3 -2 -1 -1 -1 -1 -1 0	0 -1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
160	-4 -4 -4 -3 -4 -3 -3 -3 -4 -3 -4 -4 -4 -4 -4 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
200	-6 -7 -6 -7 -6 -7 -6 -7 -6 -7 -6 -5 -6 -5 -6 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
250	-5 -7 -5 -7 -5 -7 -5 -7 -5 -7 -5 -5 -6 -6 -6 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
315	-5 -7 -5 -7 -5 -7 -5 -7 -5 -7 -5 -5 -6 -6 -6 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
400	-6 -6 -6 -6 -6 -7 -7 -7 -7 -7 -7 -7 -7 -7 -7 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
500	-7 -7 -5 -5 -7 -7 -7 -7 -7 -7 -7 -7 -7 -7 -7 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
630	-9 -7 -5 -3 -7 -7 -7 -7 -7 -7 -7 -7 -7 -7 -7 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
800	-9 -6 -6 -5 -6 -6 -6 -6 -5 -6 -6 -6 -6 -6 -6 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
1000	-9 -6 -6 -5 -6 -6 -6 -6 -5 -6 -6 -6 -6 -6 -6 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
1250	-4 -3 -1 -1 -2 -1 -1 -2 -1 -1 -2 -1 -1 -2 -1 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
1600	-1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 -1 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
2000	-3 -2 -1 -1 -2 -1 -1 -2 -1 -1 -2 -1 -1 -2 -1 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
2500	-5 -5 -3 -3 -4 -4 -4 -4 -5 -5 -5 -5 -5 -5 -5 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
3150	-2 -2 -1 -1 -3 -3 -3 -3 -4 -4 -4 -4 -4 -4 -4 0	-1 0 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
4000	0 0 2 2 2 2 2 2 2 2 2 2 2 2 2 2 0 -1 0	0 1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
5000	2 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 0	0 1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
6300	4 4 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 0	0 1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
8000	5 5 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 1 0	0 1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
10000	6 6 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 1 0	0 1 1 1 1 1 1 1 1 1 1 1 4 3 3 1	
OCTAVE	-4 -5 -4 -2 -2 -3 -2 0 0 0 1 3 -2 3 3 0 -7		
31.5	-3 -5 -5 -3 -2 -1 -3 -2 -1 -3 -4 -4 -4 -4 -4 0		
63	-3 -4 -4 -2 -1 -3 -2 -1 -3 -4 -4 -4 -4 -4 -4 0		
125	-2 -3 -3 -5 -6 -5 -6 -5 -6 -5 -6 -5 -6 -5 -6 0		
250	-5 -7 -7 -5 -5 -4 -4 -4 -4 -5 -5 -5 -5 -5 -5 0		
500	-7 -7 -5 -6 -6 -5 -6 -6 -5 -6 -6 -6 -6 -6 -6 0		
1000	-5 -5 -4 -4 -4 -4 -4 -4 -4 -5 -5 -5 -5 -5 -5 0		
2000	-3 -3 0 -1 -2 -1 -2 -1 -2 -1 -2 -1 -2 -1 -2 0		
4000	-4 -4 -1 -1 -2 -1 -2 -1 -2 -1 -2 -1 -2 -1 -2 0		
6000	5 0 2 2 1 2 1 2 1 2 1 2 1 2 1 2 1 2 1 0		
OVERALL	-4 -5 -4 -2 -2 -3 -2 0 0 0 1 3 -2 3 3 0 -7		





( FIGURE 1 OVERALL SOUND PRESSURE LEVEL (OASPL)  
**5**  
 EQUAL LEVEL CONTOURS (DB)

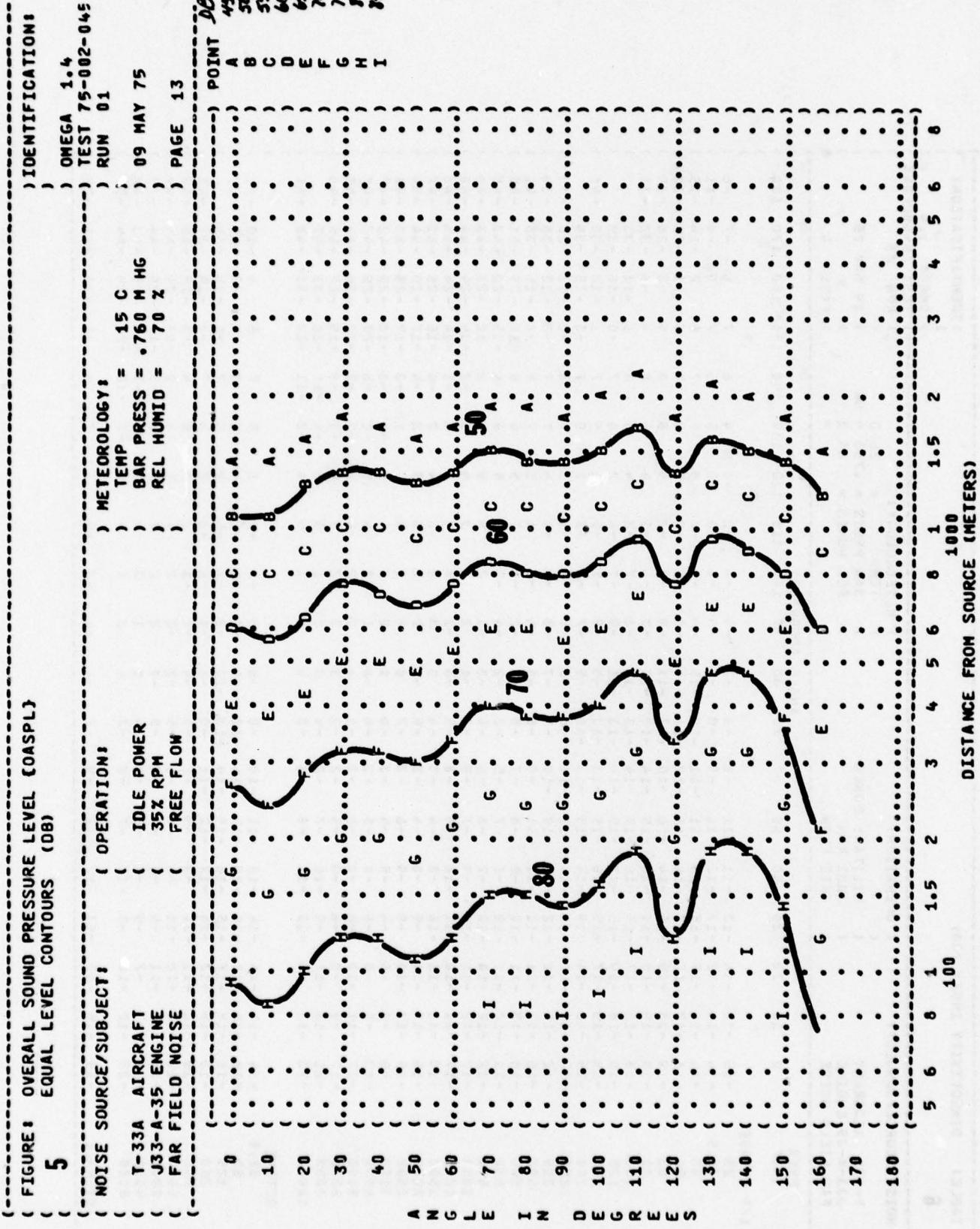


FIGURE 5  
OVERALL SOUND PRESSURE LEVEL (OASPL)  
EQUAL LEVEL CONTOURS (DB)

NOISE SOURCE/SUBJECT: T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION:  
50% RPM  
FREE FLOW

METEOROLOGY:  
TEMP = 15 C  
BAR PRESS = .760 H HG  
REL HUMID = 70 %

TEST 75-002-045  
RUN 02  
PAGE 13

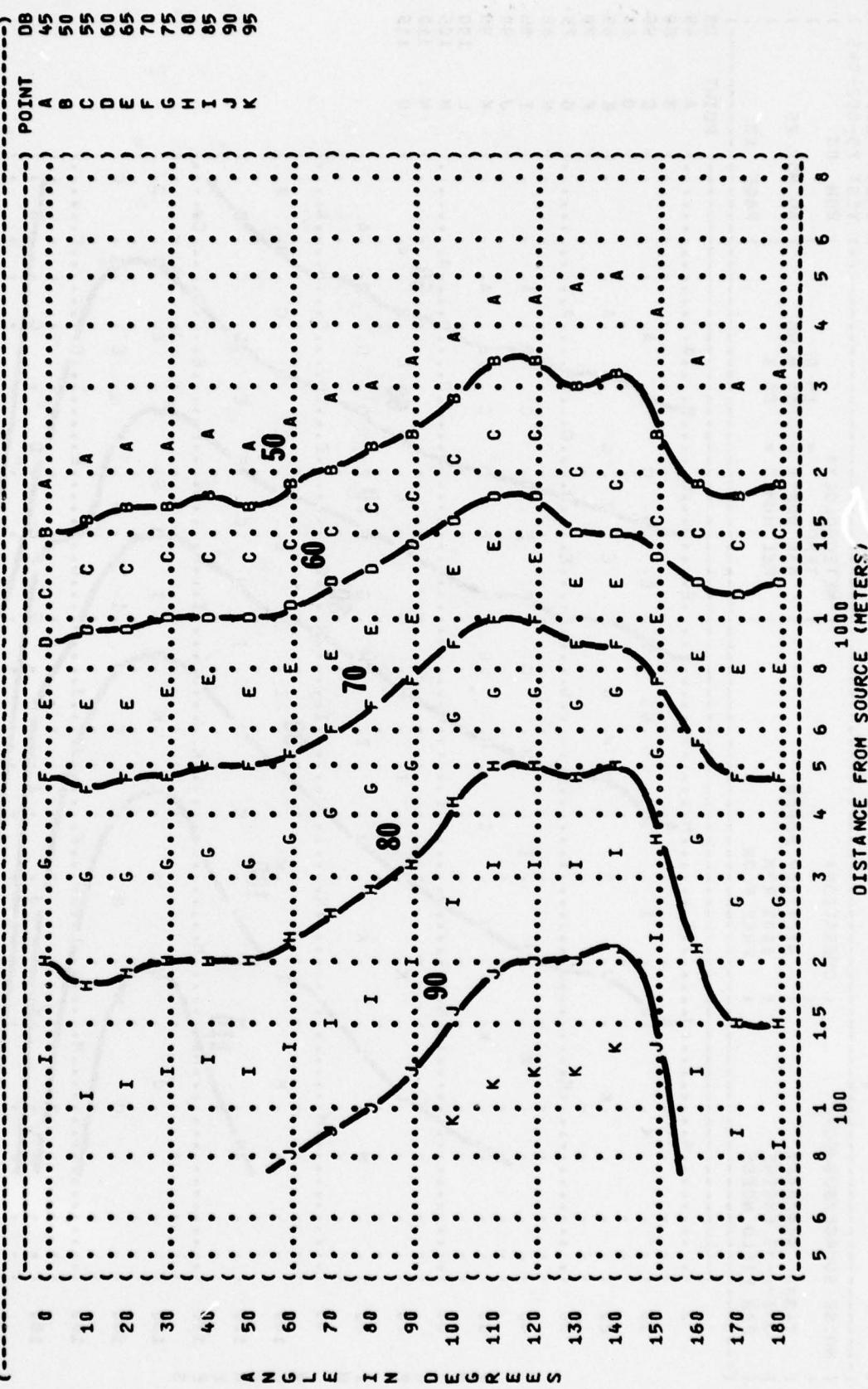


FIGURE 1 OVERALL SOUND PRESSURE LEVEL (OASPL)  
5 EQUAL LEVEL CONTOURS (DB)

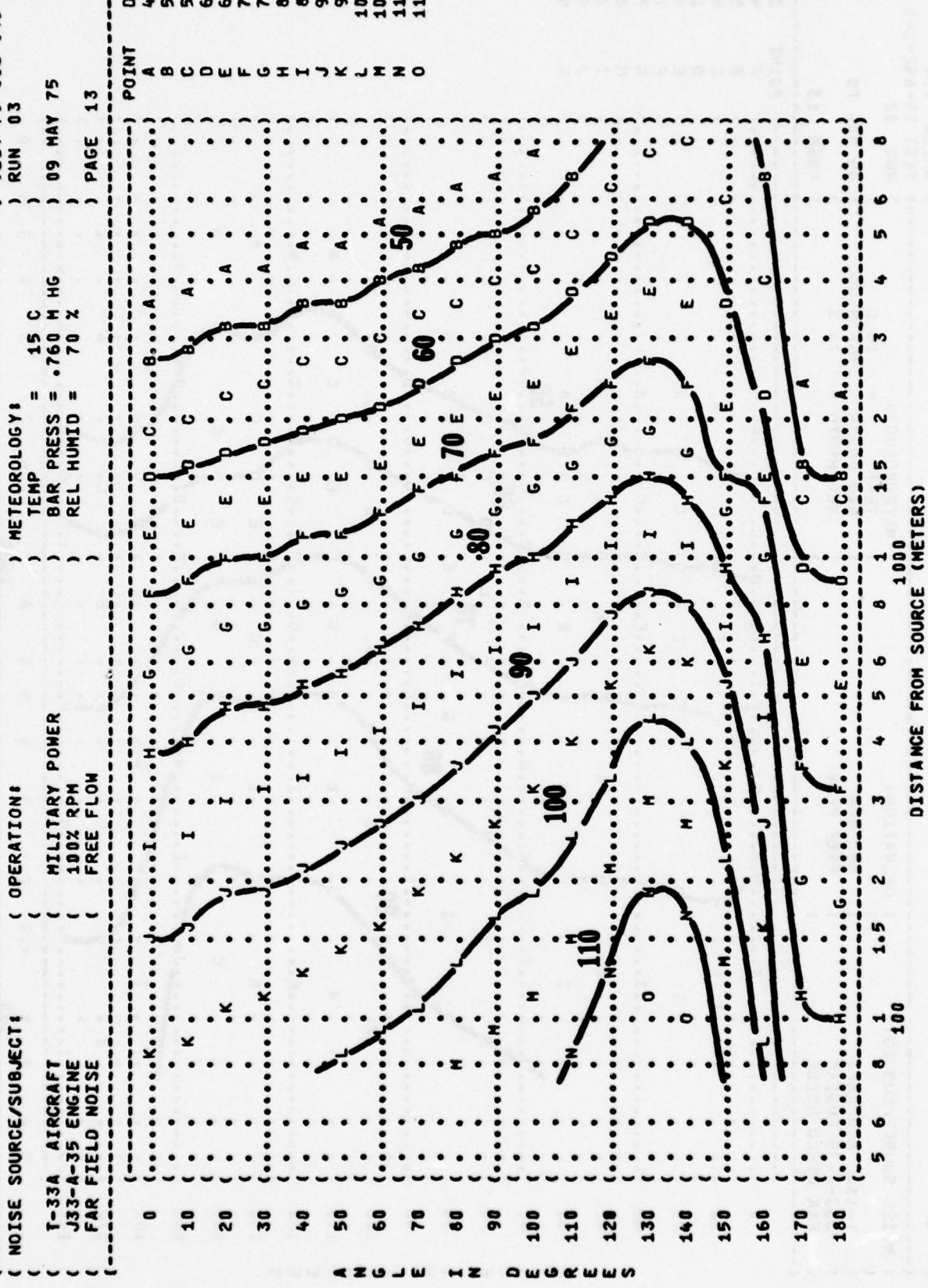


FIGURE: C-WEIGHTED OVERALL SOUND LEVEL (OASLC)  
6 EQUAL LEVEL CONTOURS (DBC)

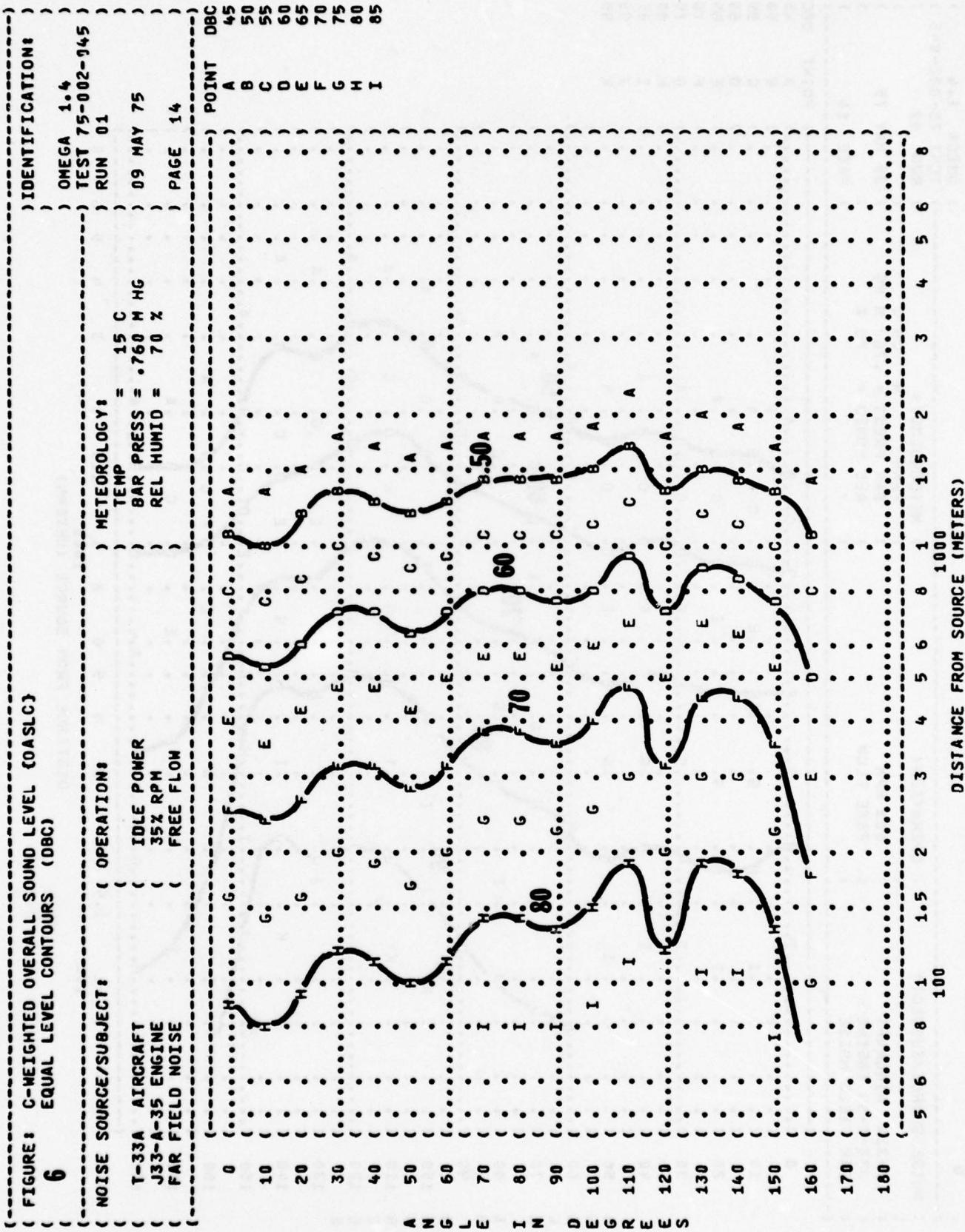


FIGURE: C-WEIGHTED OVERALL SOUND LEVEL (DBC)  
6 EQUAL LEVEL CONTOURS (DB)

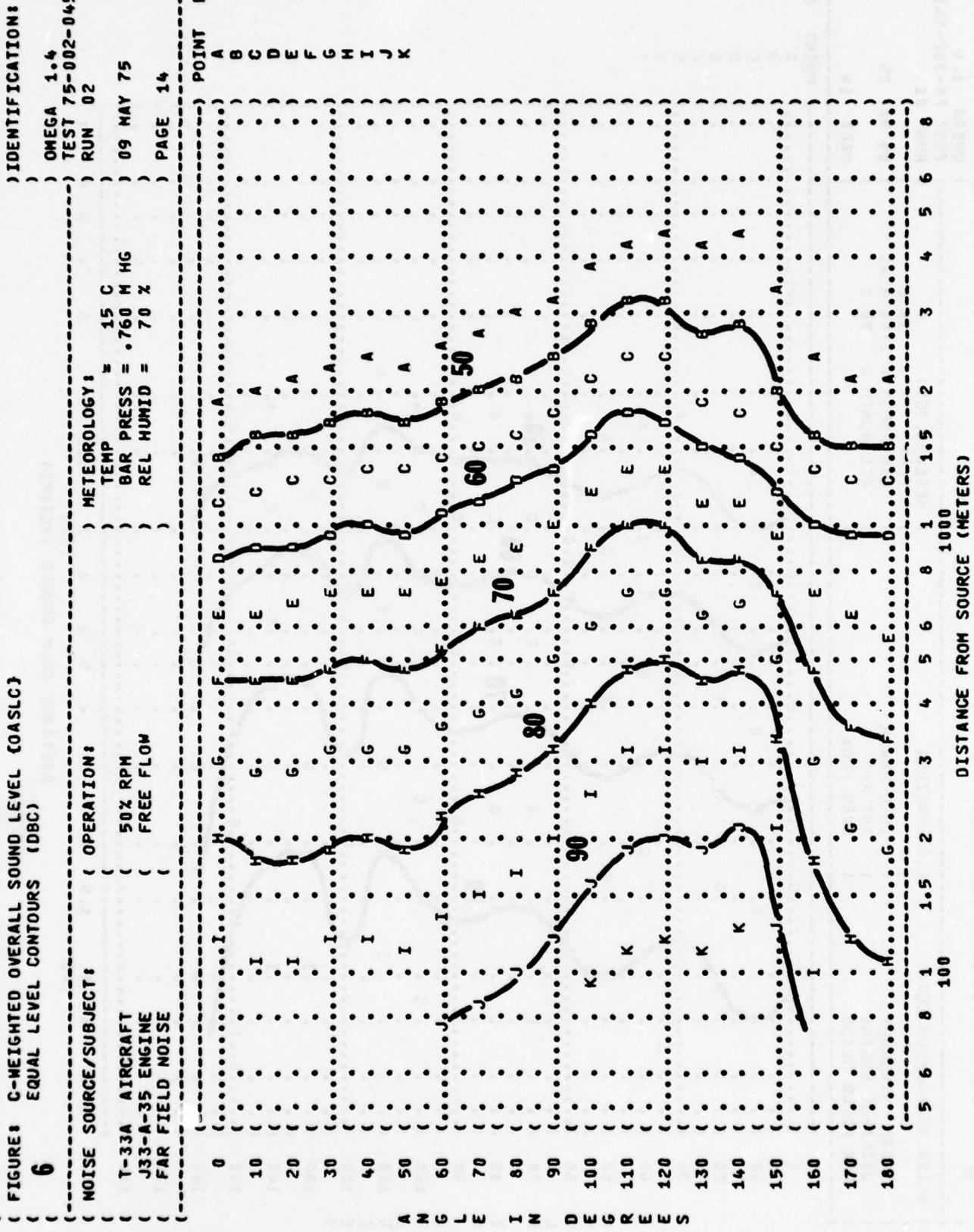


FIGURE: C-WEIGHTED OVERALL SOUND LEVEL (OASLC)  
6 EQUAL LEVEL CONTOURS (DBC)

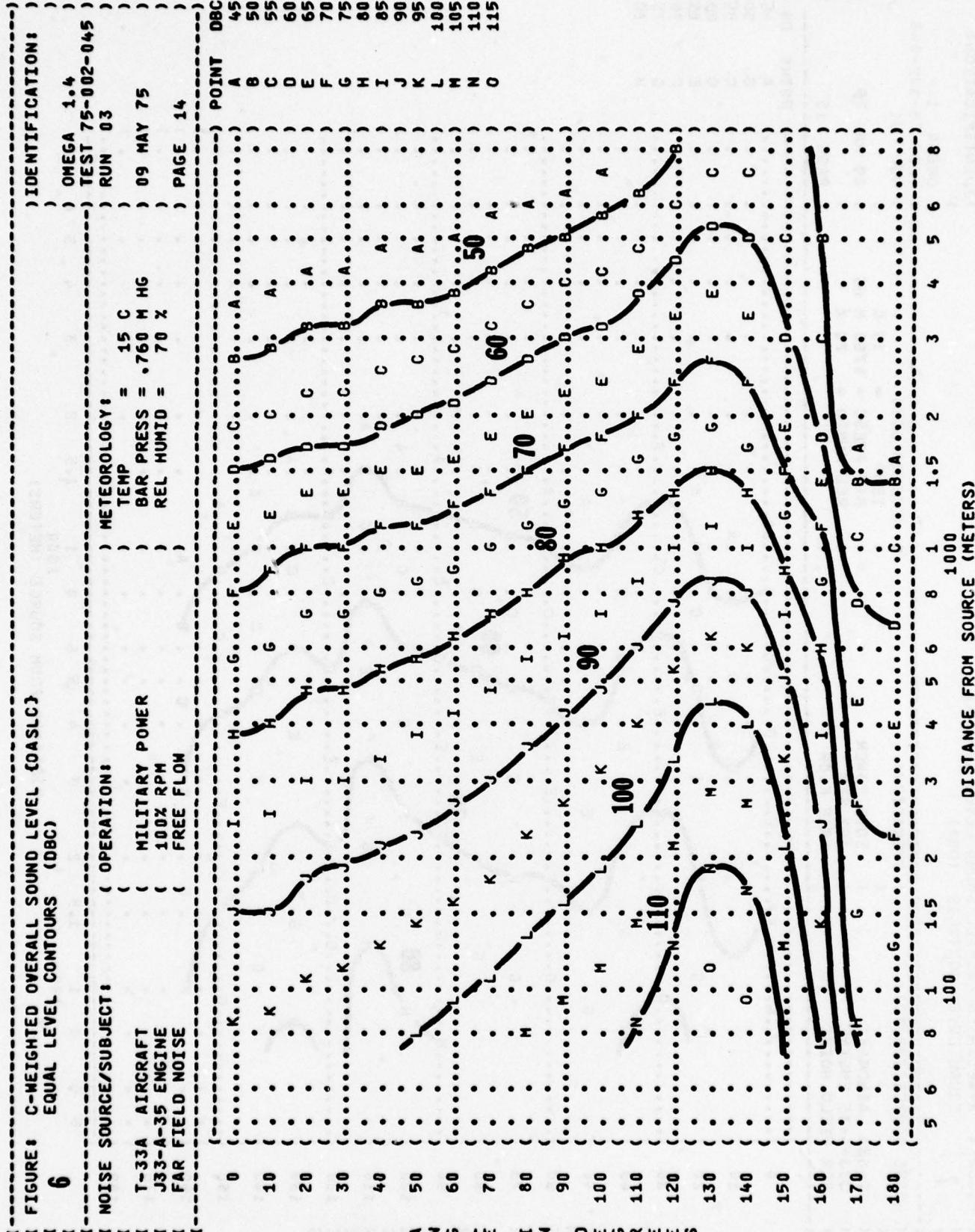


FIGURE 7 EQUAL LEVEL CONTOURS (DBA)

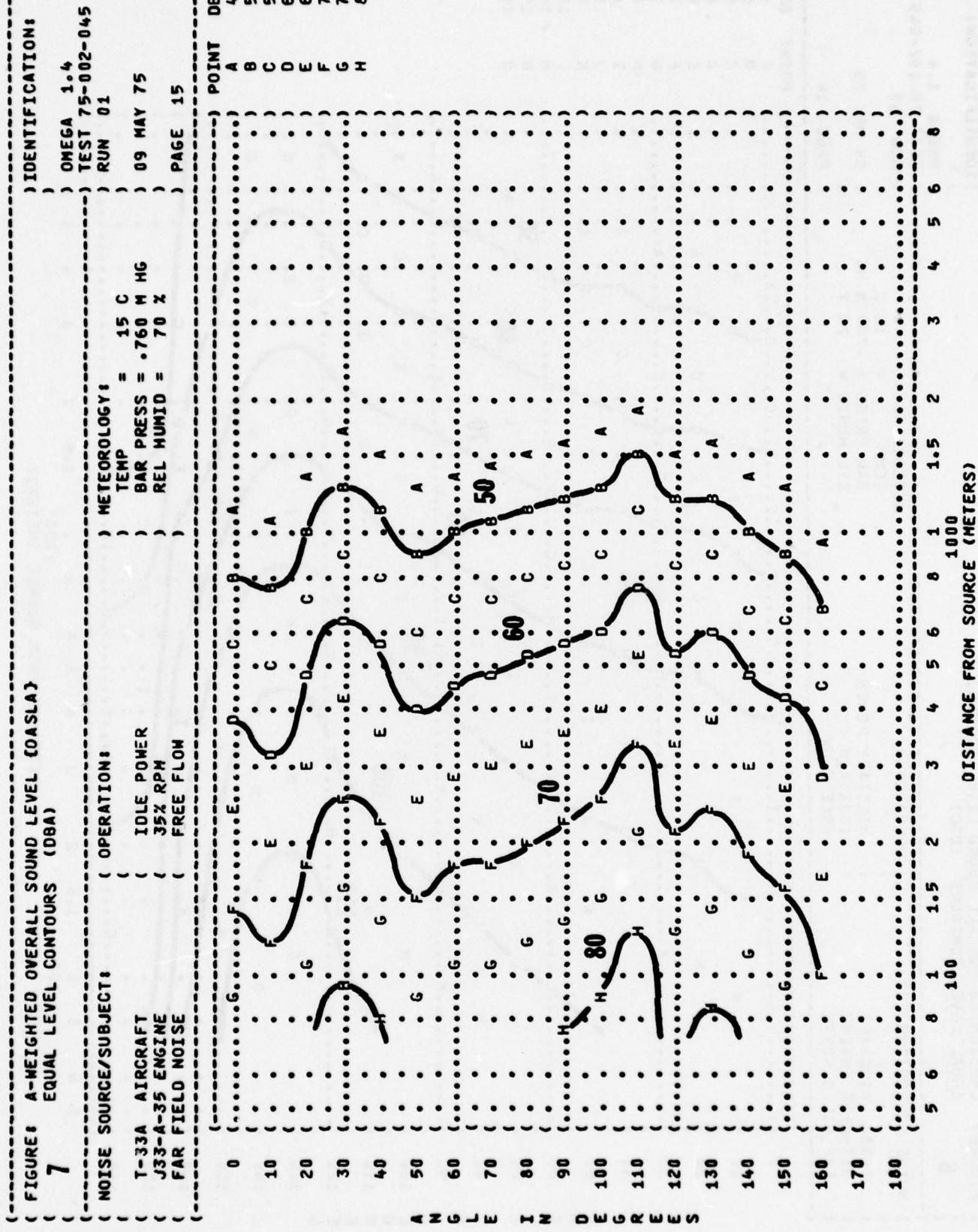


FIGURE 8 A-WEIGHTED OVERALL SOUND LEVEL (DBA)  
EQUAL LEVEL CONTOURS (DBA)

7

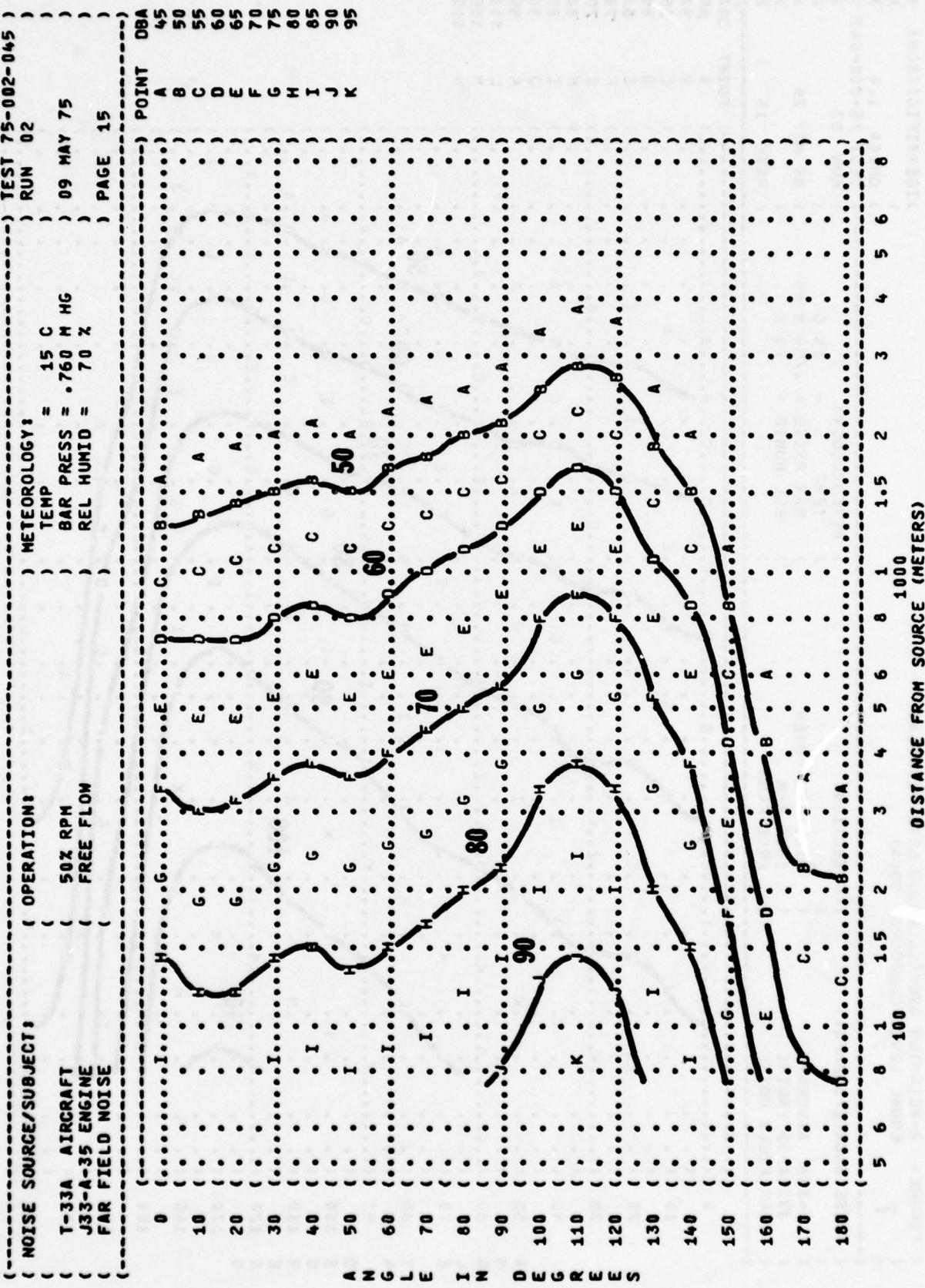


FIGURE 7  
A-WEIGHTED OVERALL SOUND LEVEL (OASLA)  
EQUAL LEVEL CONTOURS (DBA)

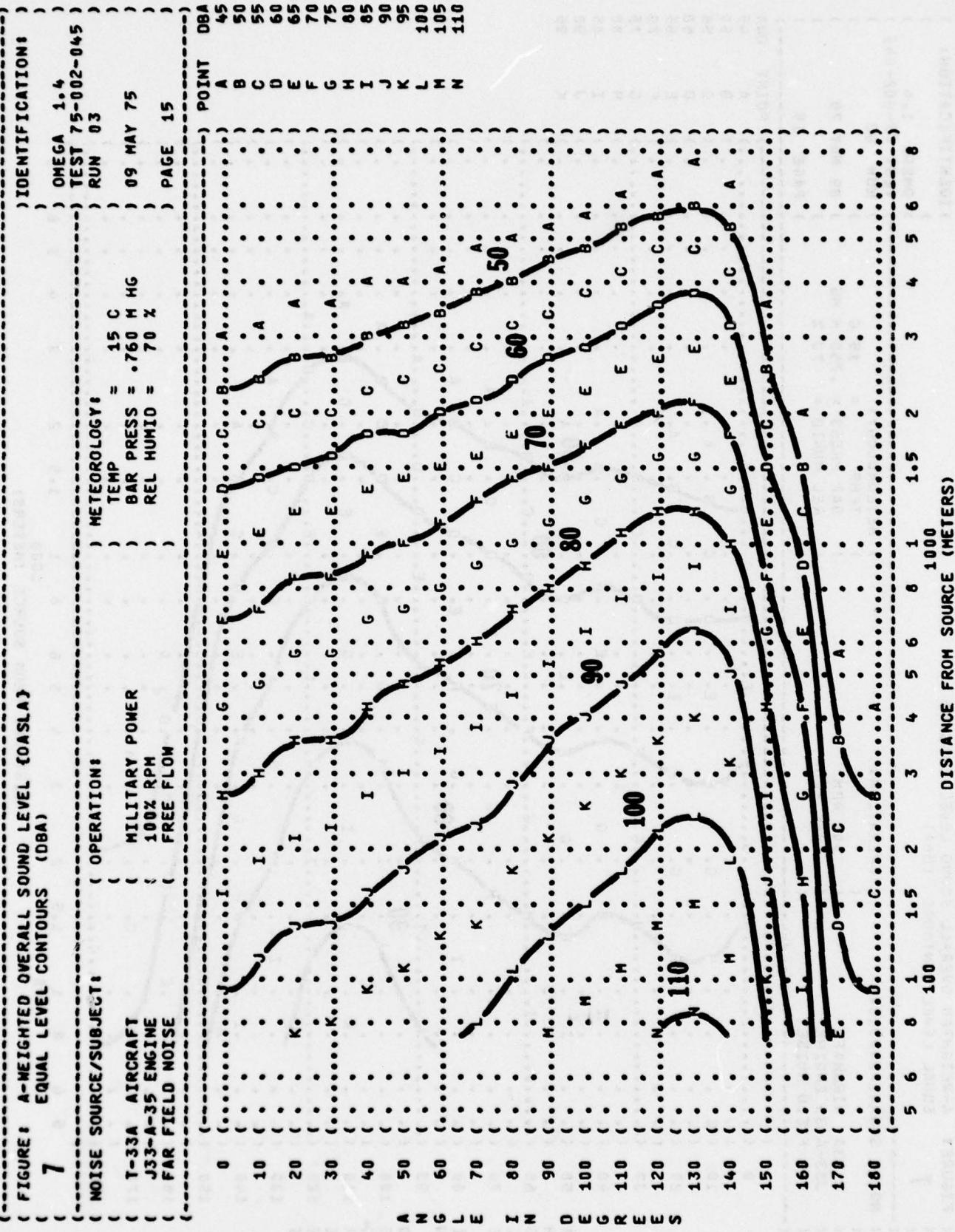


FIGURE 1 PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION (PNLT)  
8 EQUAL LEVEL CONTOURS (PNDB)

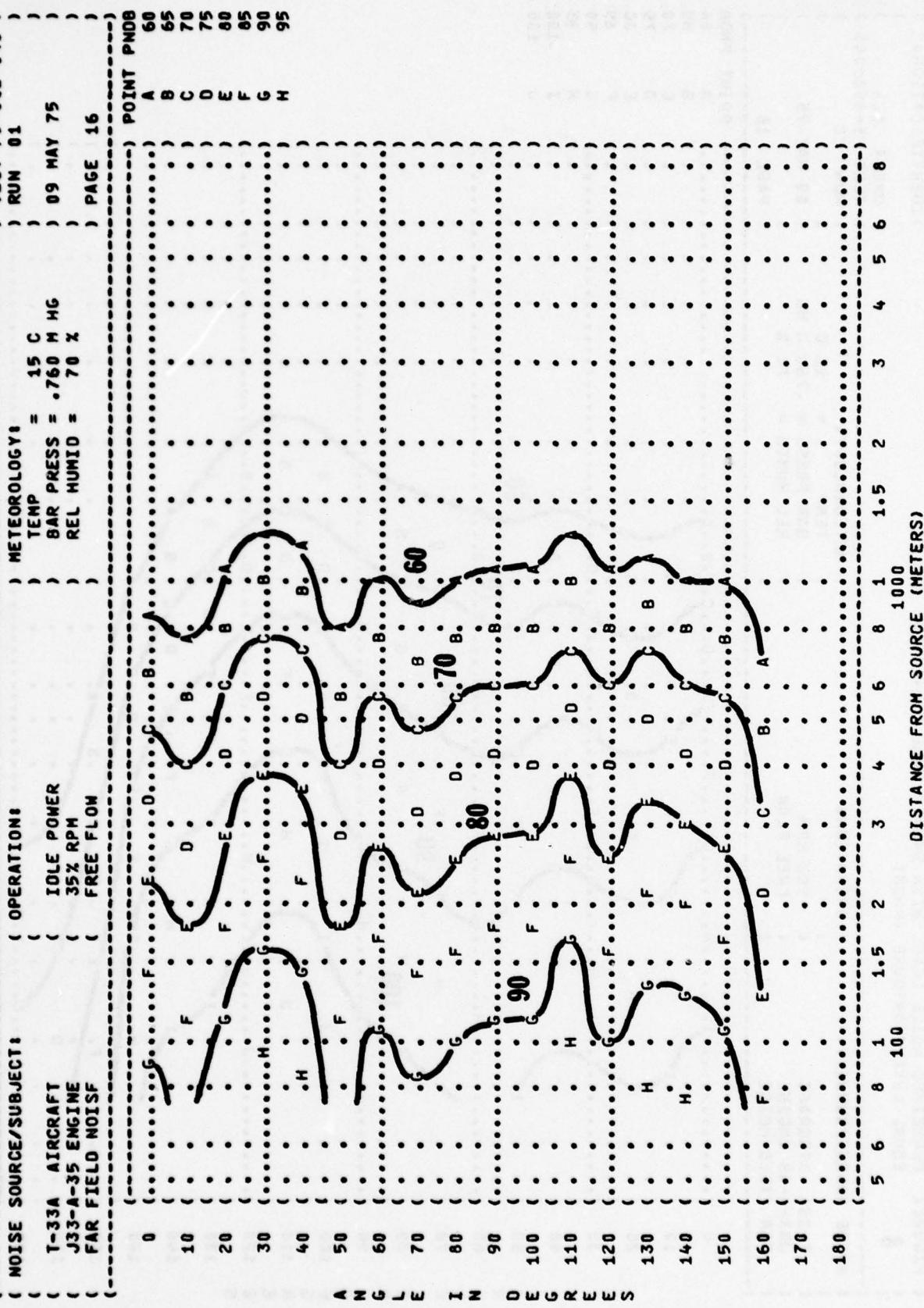


FIGURE 1 PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION (PNLT)  
8 EQUAL LEVEL CONTOURS (PNDB)

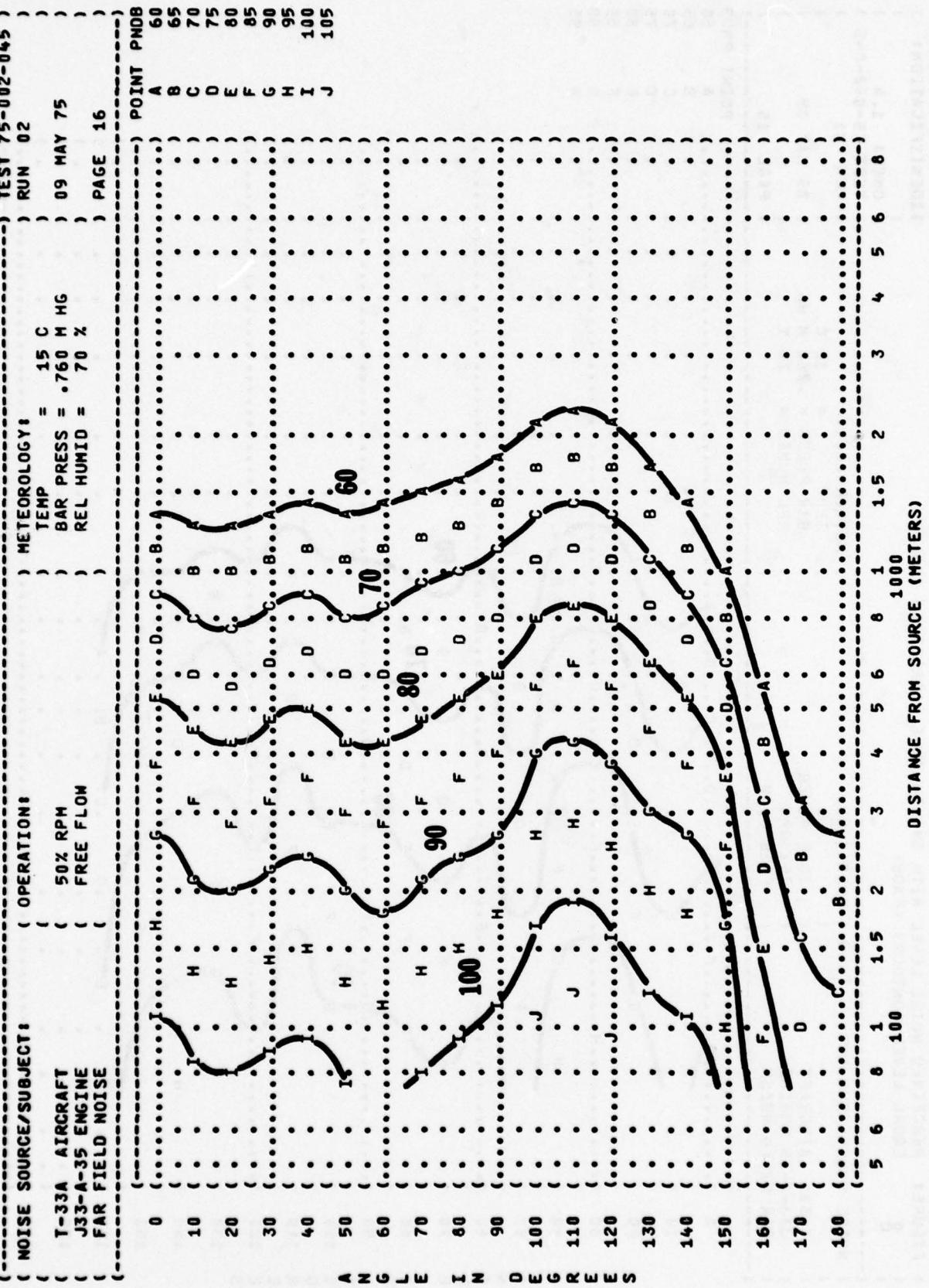
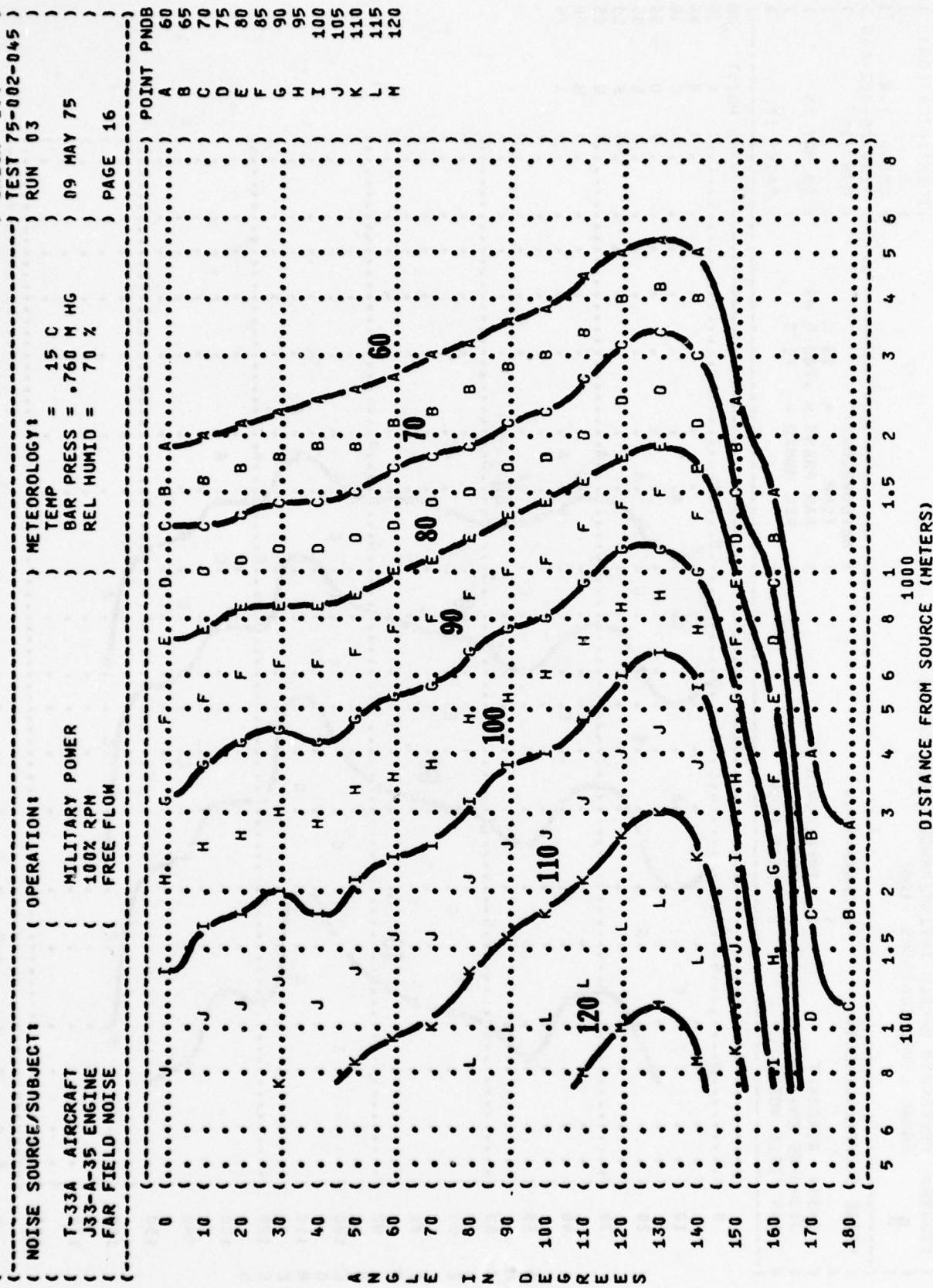
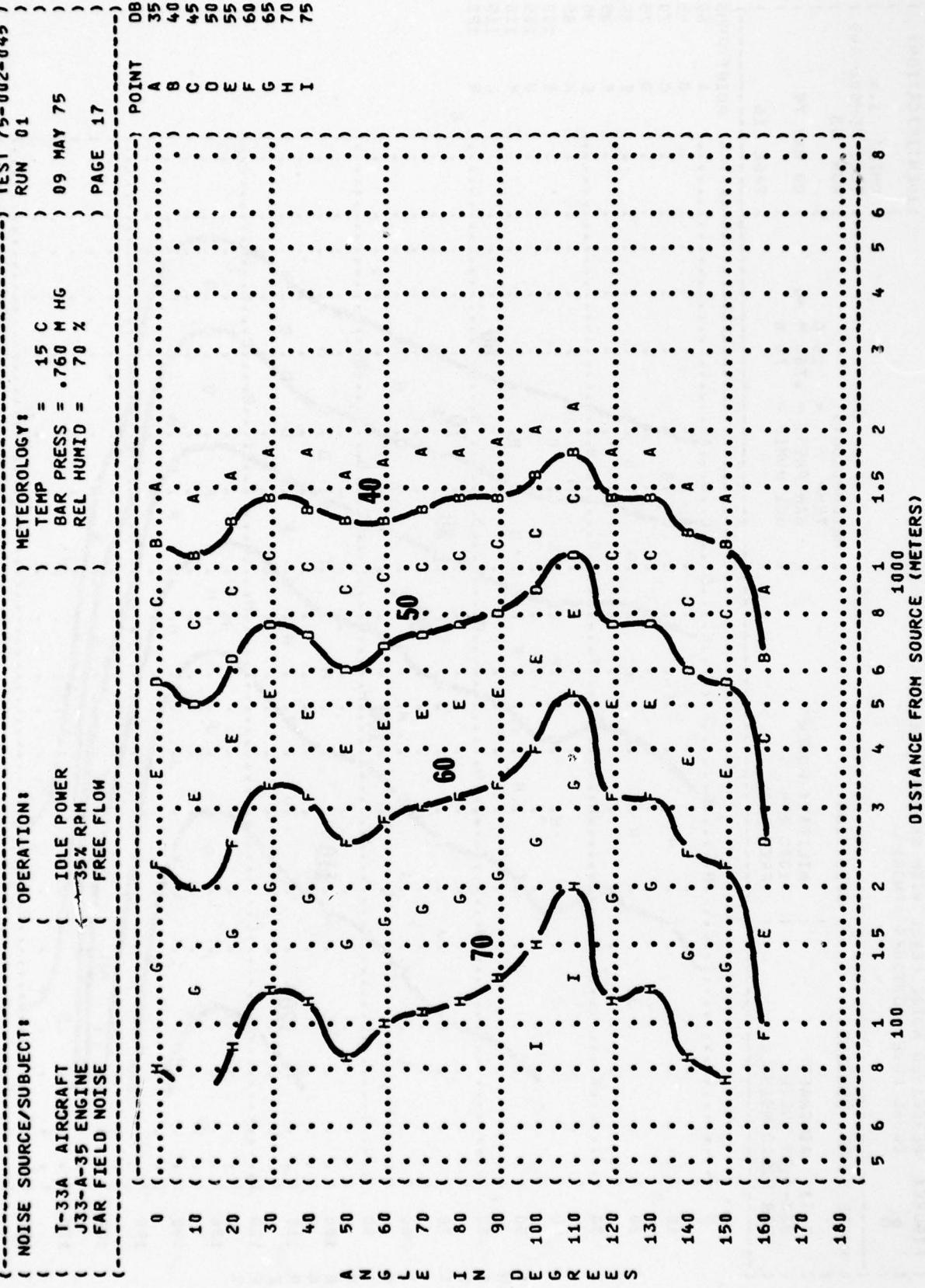


FIGURE: PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION (PNLT)  
8 EQUAL LEVEL CONTOURS (PNDB)



( FIGURE: PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)  
**9**  
 EQUAL LEVEL CONTOURS (DB)



( FIGURE 1 PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)  
 9 EQUAL LEVEL CONTOURS (DB)

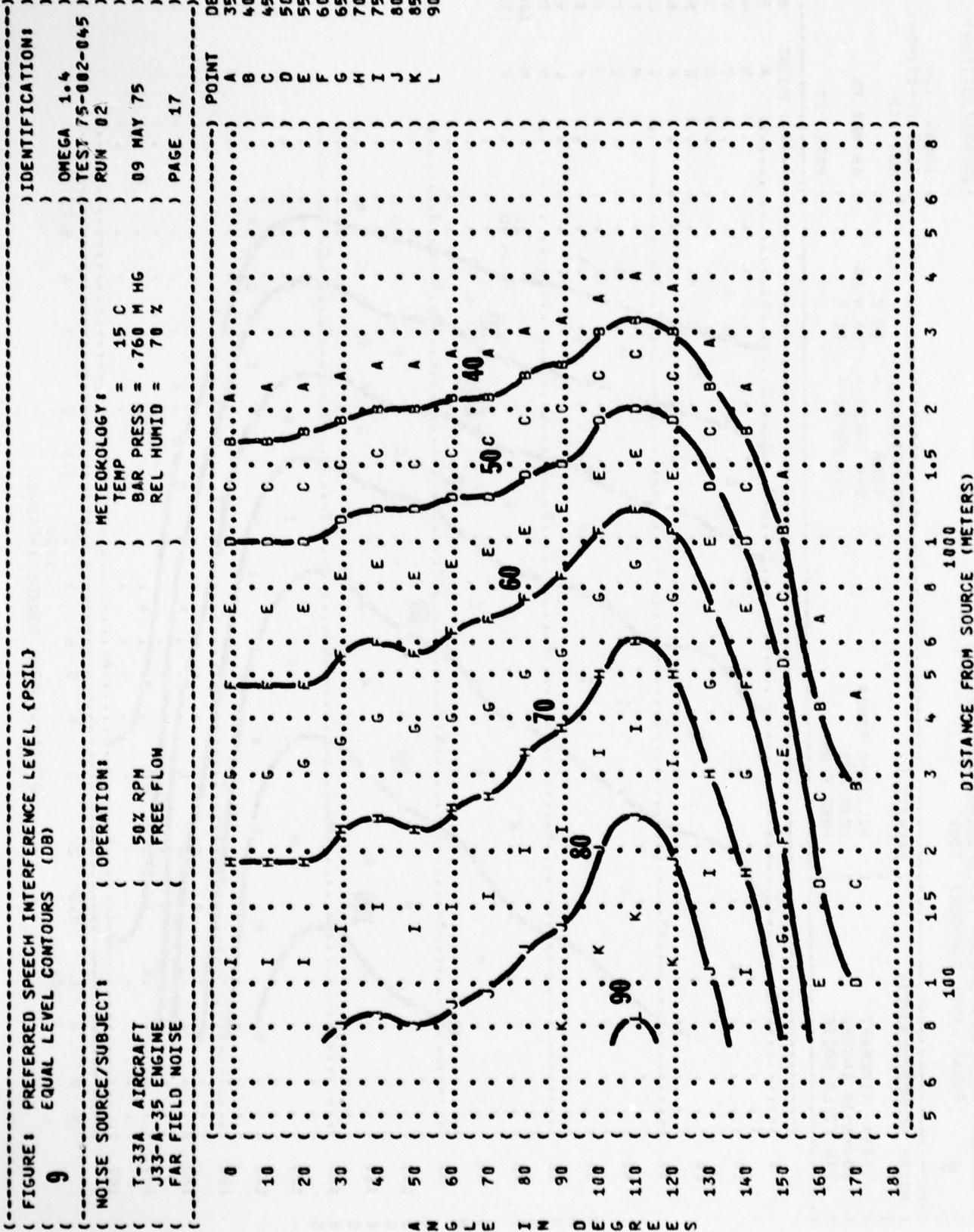
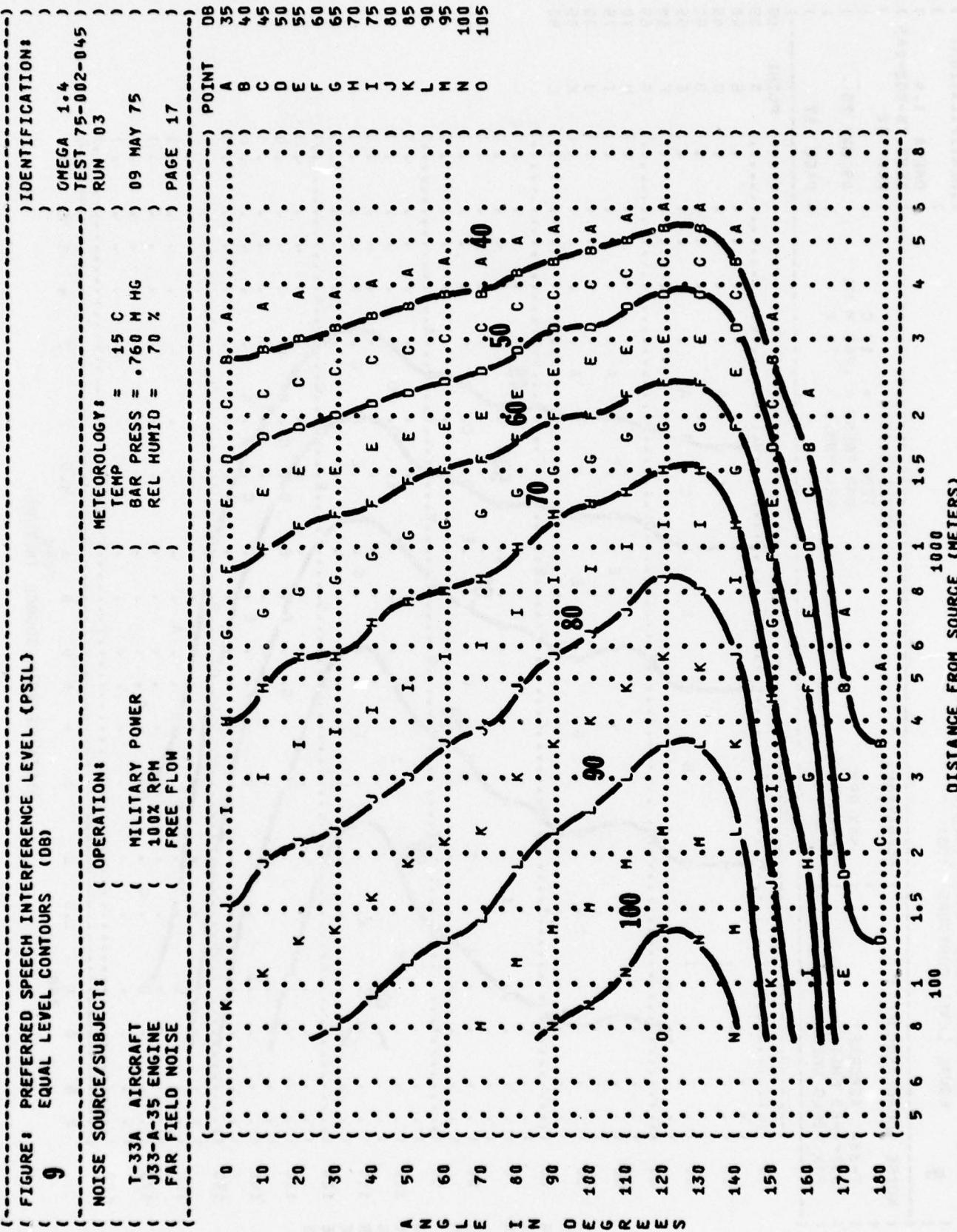


FIGURE 9 PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)



( FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73) ) IDENTIFICATION  
 EQUAL TIME CONTOURS (MINUTES)  
**10**  
 NO PROTECTION

NOISE SOURCE/SUBJECT: ( OPERATION:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

IDLE POWER  
 35% RPM  
 FREE FLOW

) METEOROLOGY:  
 ) TEMP = 15 C  
 ) BAR PRESS = .760 MM HG  
 ) REL HUMID = 70 %  
 ) PAGE 7

POINT MIN

A 960

B 480

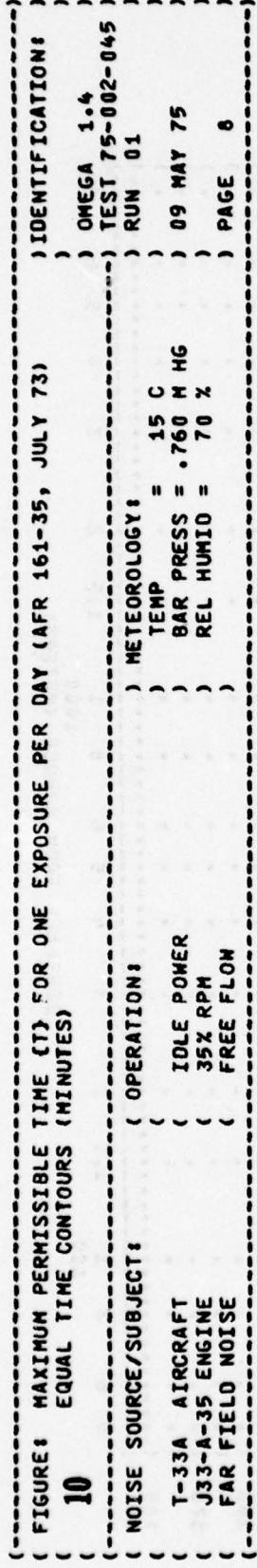
0  
 10  
 20  
 30 A  
 40  
 50  
 60  
 70  
 80  
 90  
 D  
 E 100  
 G  
 R 110  
 E 120  
 S  
 130  
 140  
 150  
 160  
 170  
 180

480

A

A

5 6 8 1 1.5 2 3 4 5 6 8 1 1.5 2 3 4 5 6 8  
 100 DISTANCE FROM SOURCE (METERS)



NOISE SOURCE/SUBJECT: T-33A AIRCRAFT J33-A-35 ENGINE FAR FIELD NOISE

OPERATION: IDLE POWER 35% RPM FREE FLOW

METEOROLOGY: TEMP = 15 C BAR PRESS = .760 Hg REL HUMID = 70 %

TEST 75-002-045 RUN 01 OMEGA 1.4 PAGE 6

PERSONNEL MAY BE EXPOSED UP TO 960 MINUTES PER DAY

A 50° N 60° G L E 70° I N D E S T A N C E S F R O M S O U R C E E Q U A L T O O R G R E A T E R T H A N 75 M E T E R S  
F O R A L L A N G L E S E V A L U A T E D ( I N D I C A T E D B Y < AT L E F T )

UNDER THE FOLLOWING EAR PROTECTION CONDITIONS:

MINIMUM QPL EAR MUFFS

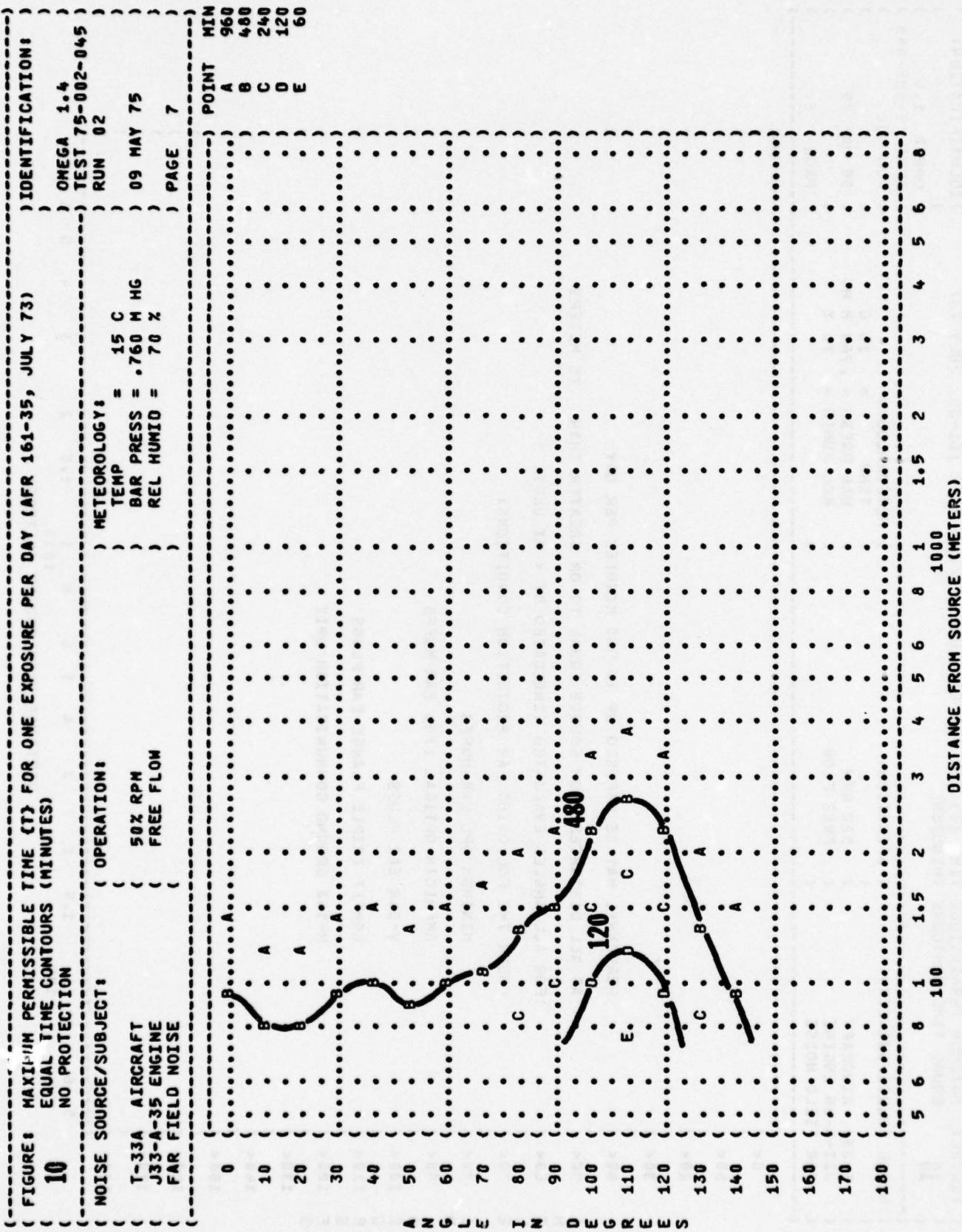
AMERICAN OPTICAL 1700 EAR MUFFS

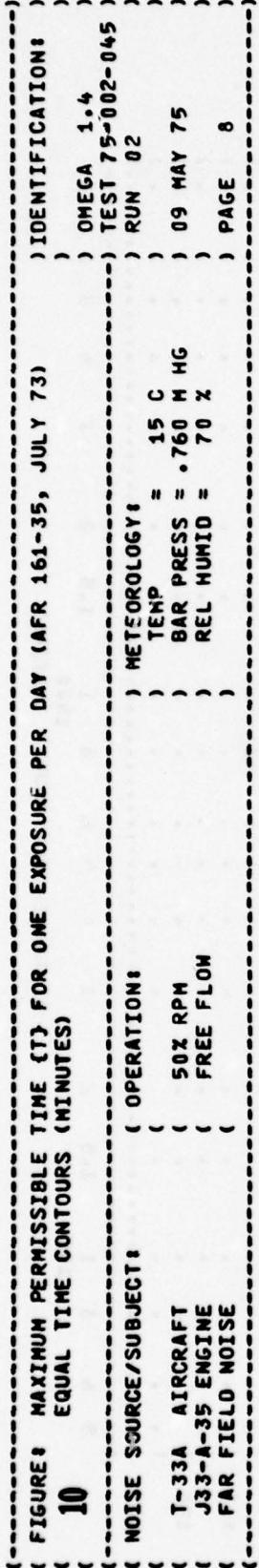
V-51R EAR PLUGS

COMFIT TRIPLE FLANGE EAR PLUGS

H-133 GROUND COMMUNICATION UNIT

5 6 8 1 1.5 2 3 4 5 6 8 1 1.5 2 3 4 5 6 8  
100 1000 DISTANCE FROM SOURCE (METERS)





NOISE SOURCE/SUBJECT: ( OPERATION: ) METEOROLOGY: )

T-33A AIRCRAFT ( 50% RPM ) TEMP = 15 C  
J33-A-35 ENGINE ( FREE FLOW ) BAR PRESS = .760 Hg  
FAR FIELD NOISE ( ) REL HUMID = 70 % ) 09 MAY 75  
( ) PAGE 8 )

PERSONNEL MAY BE EXPOSED UP TO 960 MINUTES PER DAY  
AT ALL DISTANCES FROM SOURCE EQUAL TO OR GREATER THAN 75 METERS  
FOR ALL ANGLES EVALUATED (INDICATED BY < AT LEFT)  
UNDER THE FOLLOWING EAR PROTECTION CONDITIONS:

MINIMUM QPL EAR MUFFS  
AMERICAN OPTICAL 1700 EAR MUFFS

V-51R EAR PLUGS

COMFIT TRIPLE FLANGE EAR PLUGS

H-133 GROUND COMMUNICATION UNIT

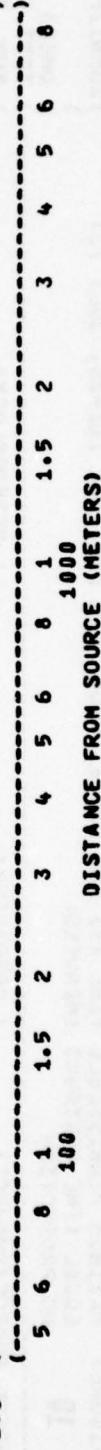
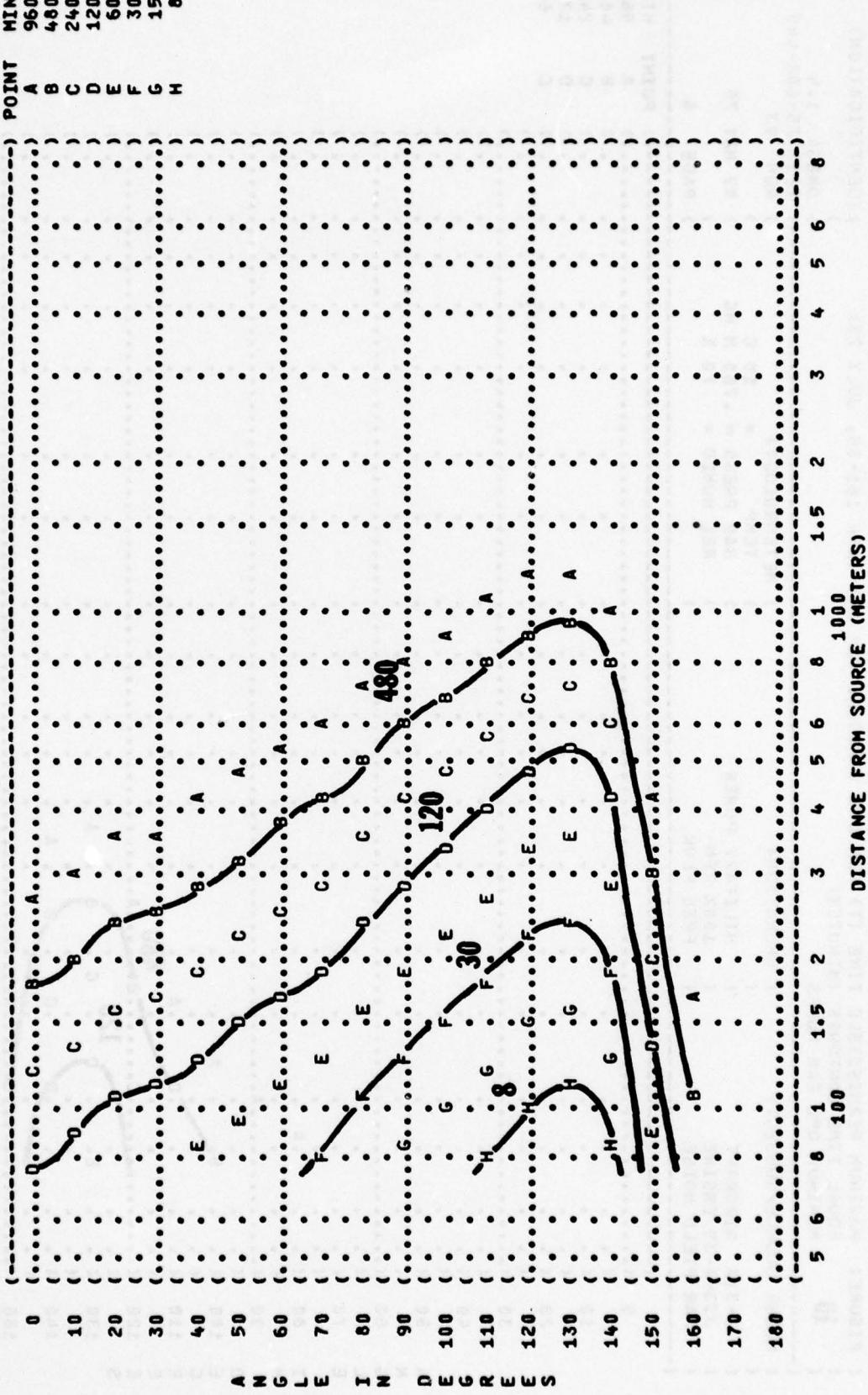


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)

**10** EQUAL TIME CONTOURS (MINUTES)

**10** NO PROTECTION

NOISE SOURCE/SUBJECT:	OPERATION:	METEOROLOGY:	POINT	MIN
T-33A AIRCRAFT J33-A-35 ENGINE FAR FIELD NOISE	MILITARY POWER 100X RPM FREE FLOW	TEMP = 15 C BAR PRESS = .760 MM HG REL HUMID = 70 %	A	960
			B	480
			C	240
			D	120
			E	60
			F	30
			G	15
			H	6



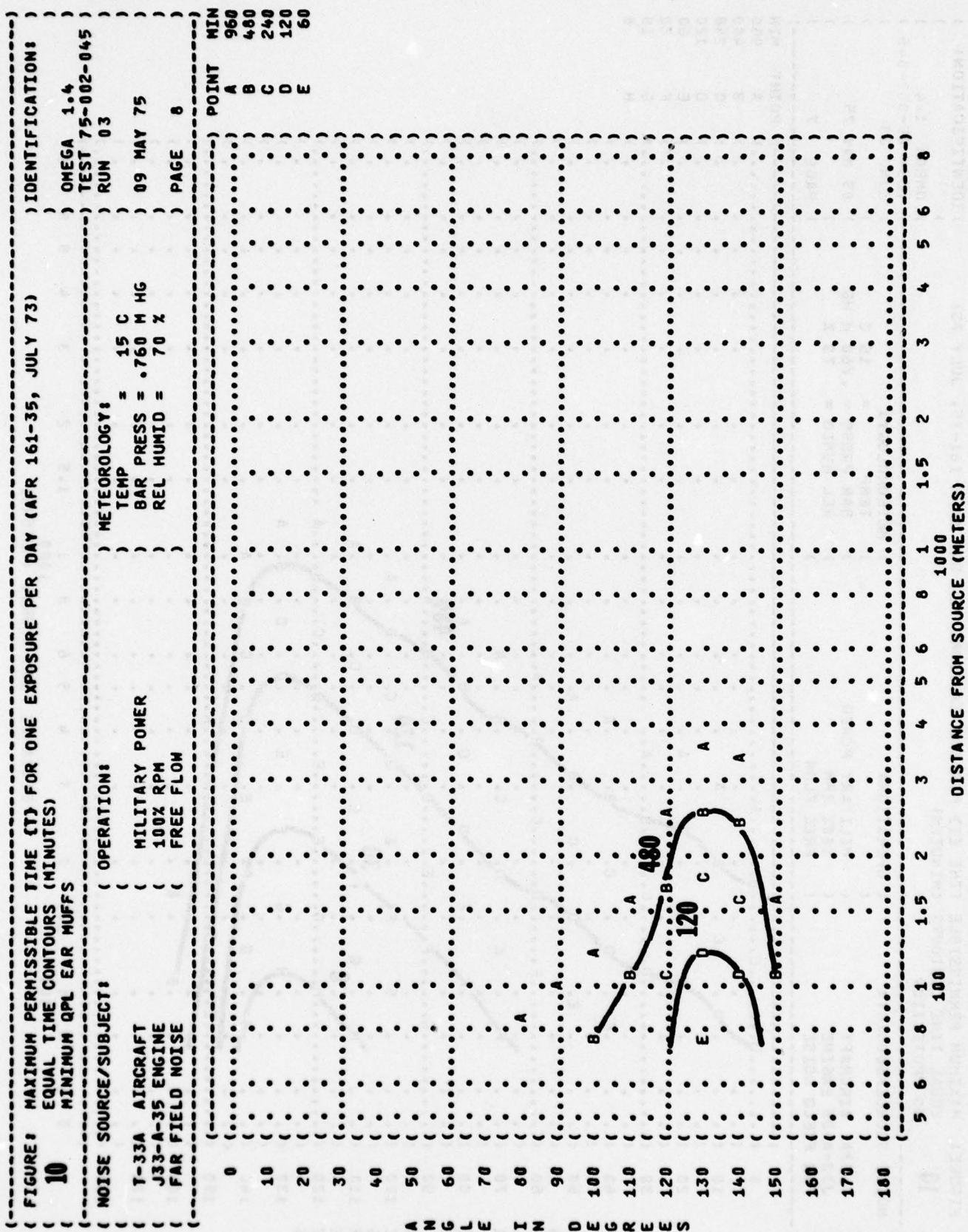


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)  
 EQUAL TIME CONTOURS (MINUTES)  
**10**  
 AMERICAN OPTICAL 1700 EAR MUFFS

NOISE SOURCE/SUBJECT: **T-33A AIRCRAFT**  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

OPERATION: **MILITARY POWER**  
 100X RPM  
 FREE FLOW

METEOROLOGY:  
 TEMP = 15 C  
 BAR PRESS = .760 Hg  
 REL HUMID = 70 %

PAGE 9

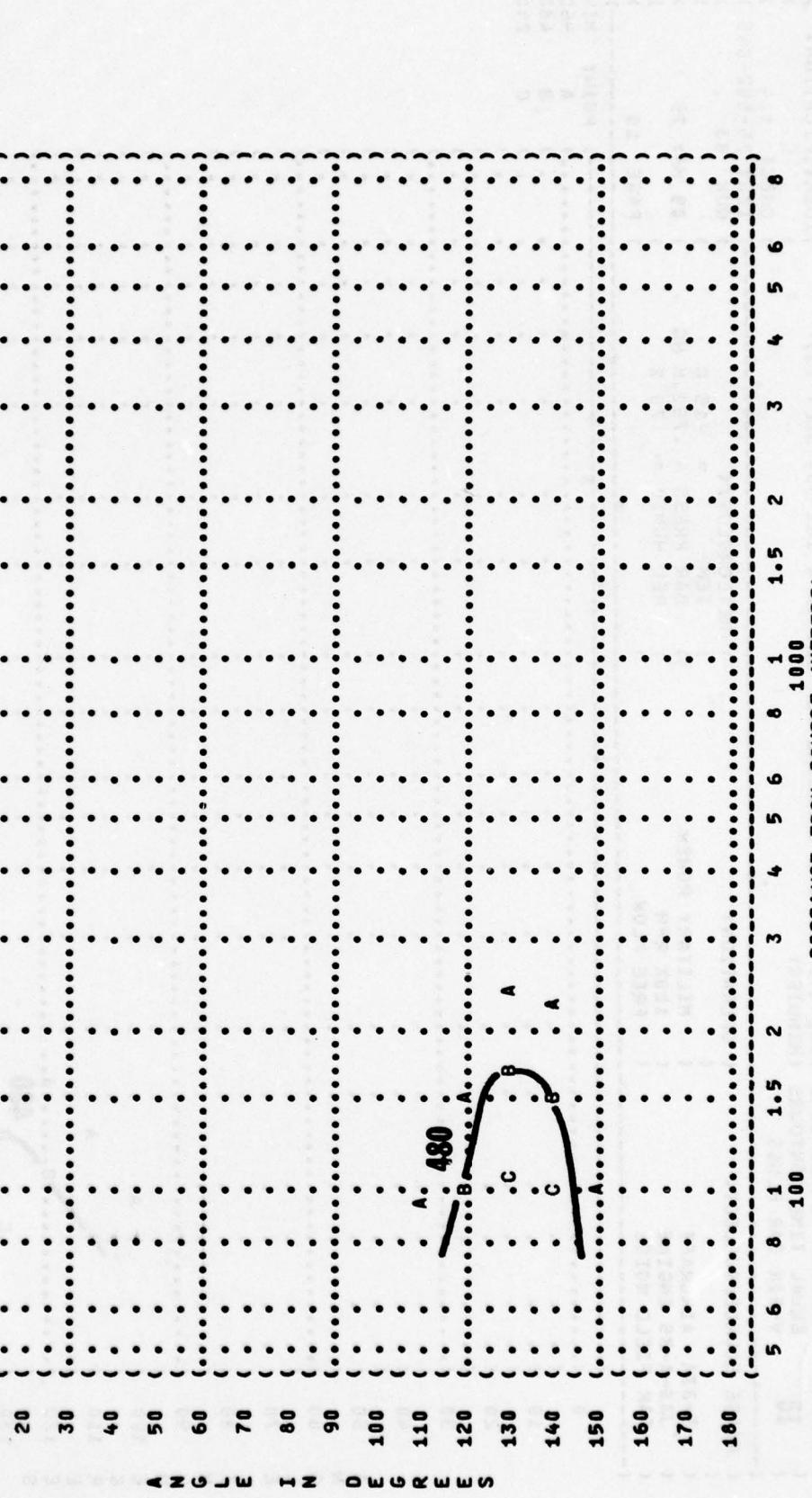


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)  
10 EQUAL TIME CONTOURS (MINUTES)  
V-SIR EAR PLUGS

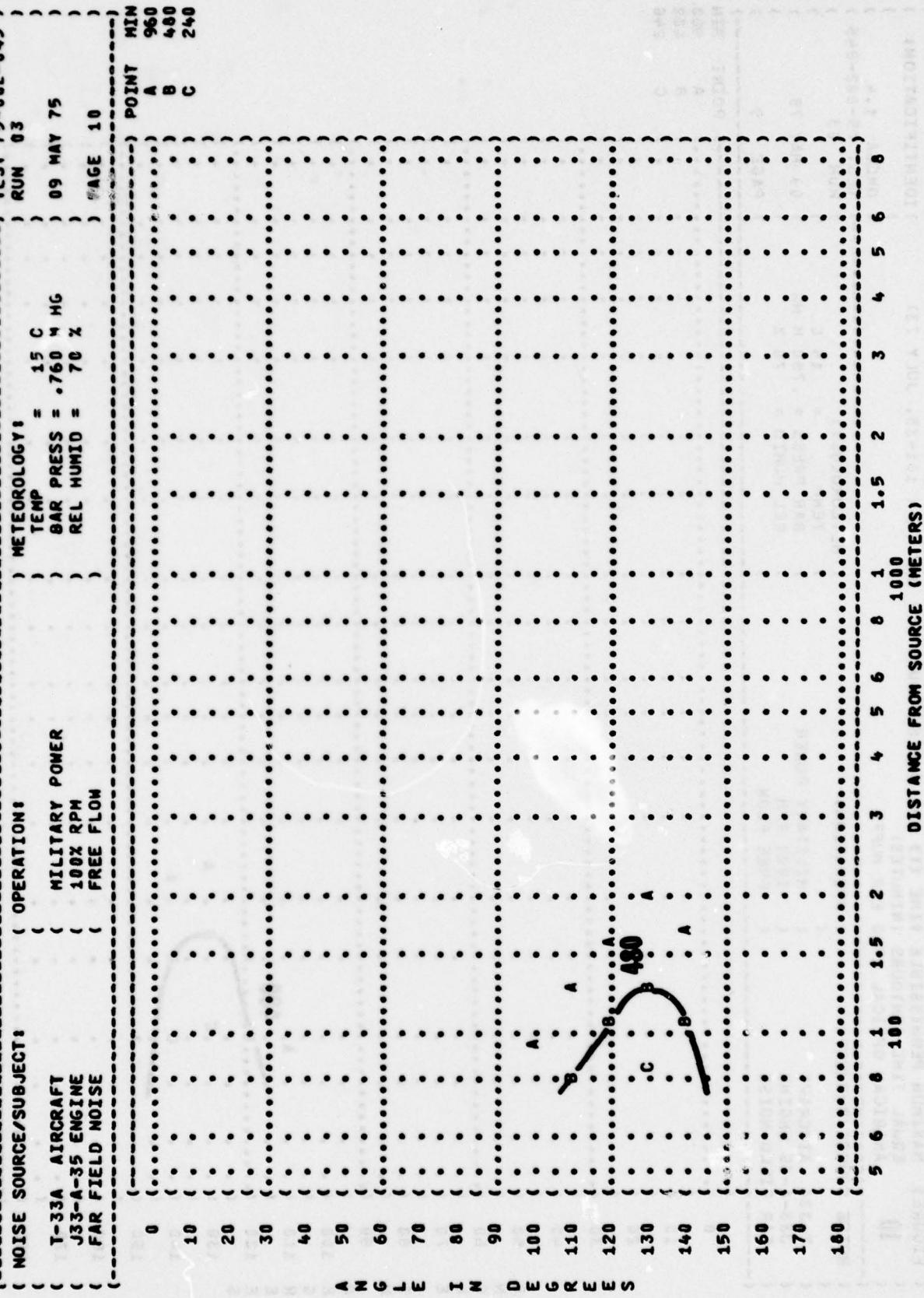


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)  
**10** EQUAL TIME CONTOURS (MINUTES)  
 COMFIT TRIPLE FLANGE EAR PLUGS

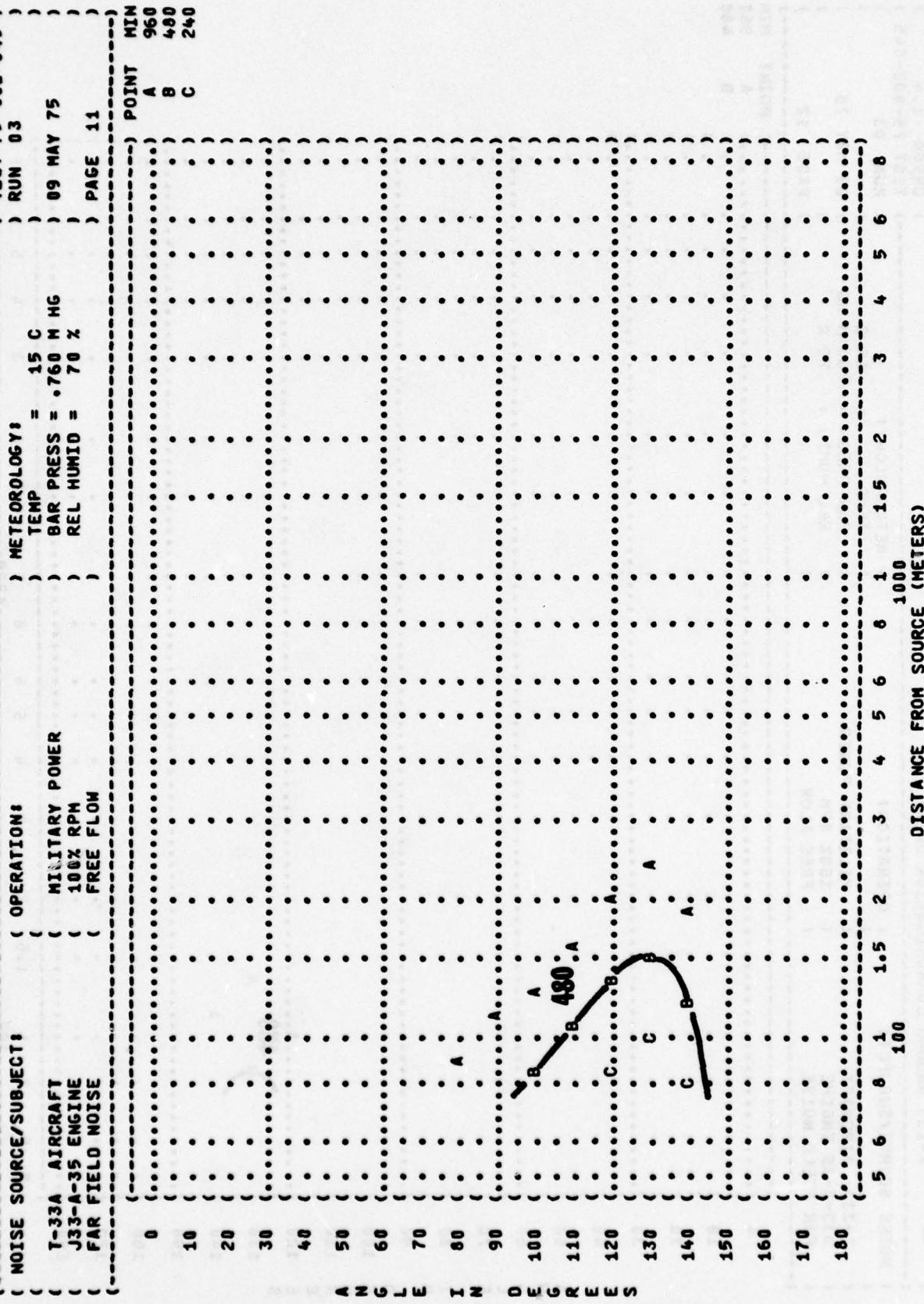


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)

10 EQUAL TIME CONTOURS (MINUTES)

H-133 GROUND COMMUNICATION UNIT

NOISE SOURCE/SUBJECT: T-33A AIRCRAFT

J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION: MILITARY POWER

100% RPM  
FREE FLOW

METEOROLOGY: TEMP = 15 C

BAR PRESS = .760 M HG

REL HUMID = 70%

PAGE 12

POINT MIN

A 960

B 480

0

10

20

30

40

50

60

70

80

90

100

110

120

130

140

150

160

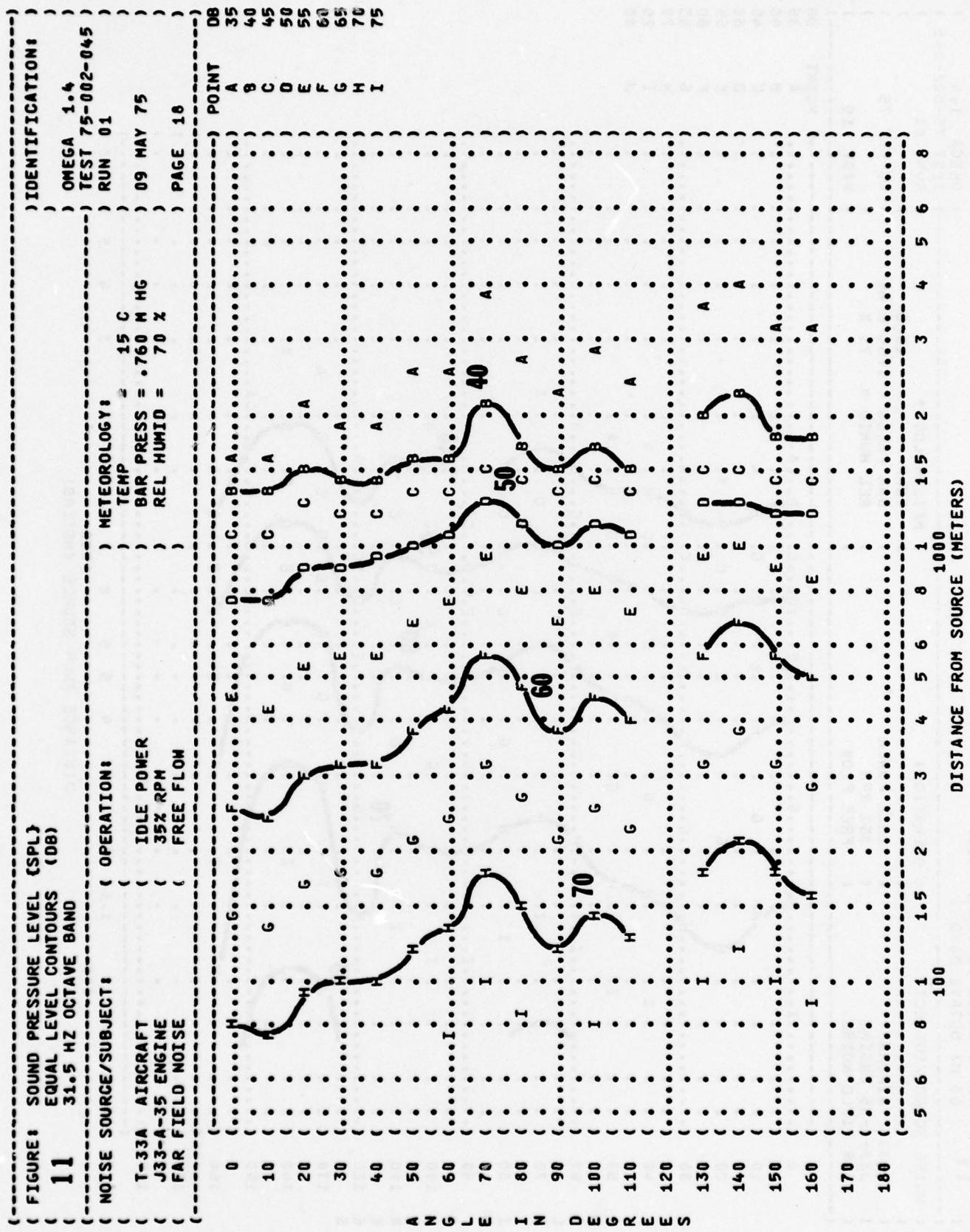
170

180

5 6 8 1 1.5 2 3 4 5 6 8 1 1.5 2 3 4 5 6

1000 DISTANCE FROM SOURCE (METERS)

IDENTIFICATION: OMEGA 1.4  
TEST 75-002-045  
RUN 03



( FIGURE: SOUND PRESSURE LEVEL (SPL)  
 11 EQUAL LEVEL CONTOURS (DB)  
 63 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

IDLE POWER  
 35% RPM  
 FREE FLOW

OPERATION:  
 ) IDENTIFICATION:  
 ) OMEGA 1.4  
 ) TEST 75-002-045  
 ) RUN 01  
 ) METEOROLOGY:  
 ) TEMP = 15 C  
 ) BAR PRESS = .760 M HG  
 ) REL HUMID = 70 %  
 ) PAGE 19

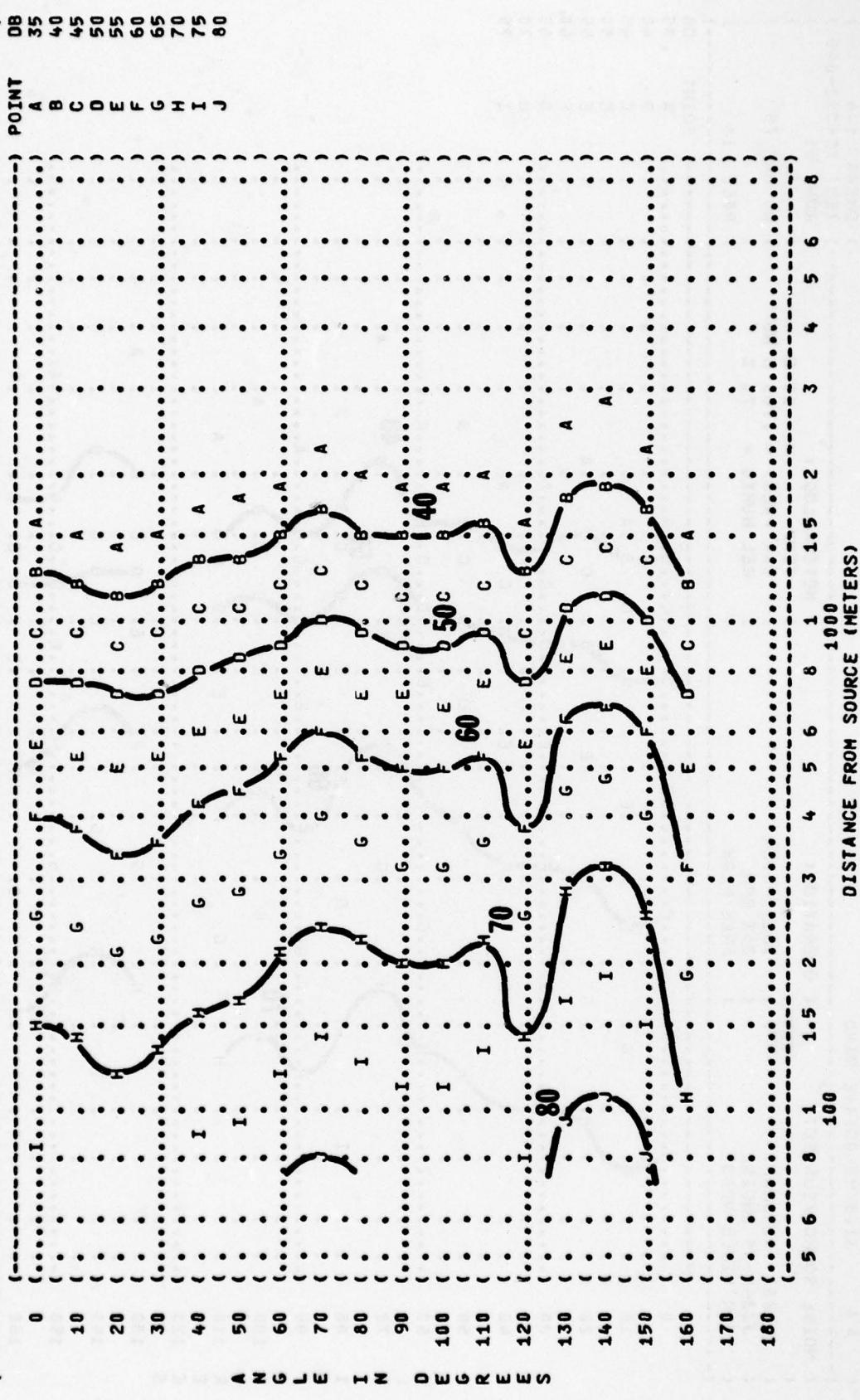


FIGURE: SOUND PRESSURE LEVEL (SPL)  
**11**  
 EQUAL LEVEL CONTOURS  
 125 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:  
**T-33A AIRCRAFT**  
**J33-A-35 ENGINE**  
**FAR FIELD NOISE**

OPERATION:  
 IDLE POWER  
 352 RPM  
 FREE FLOW

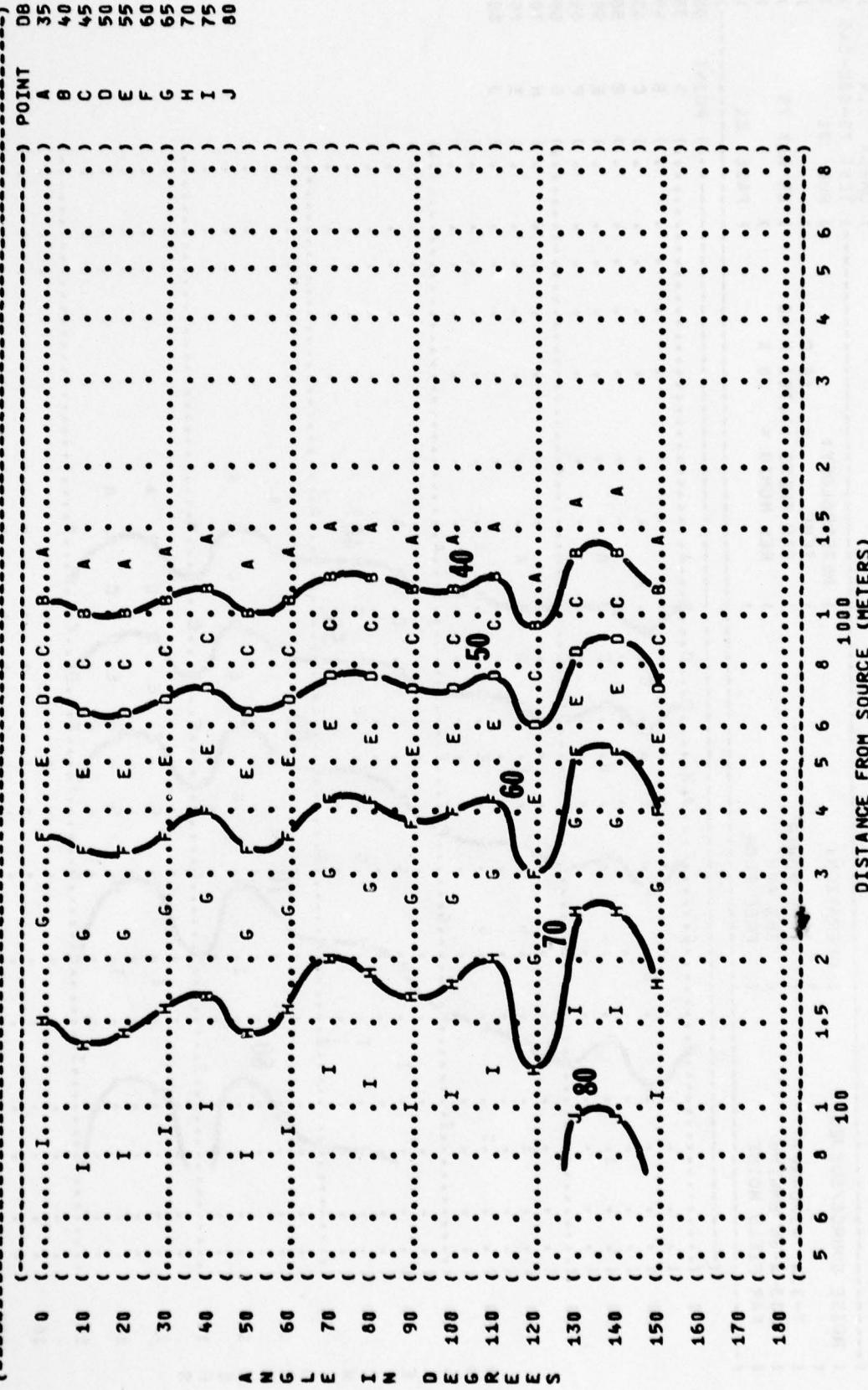
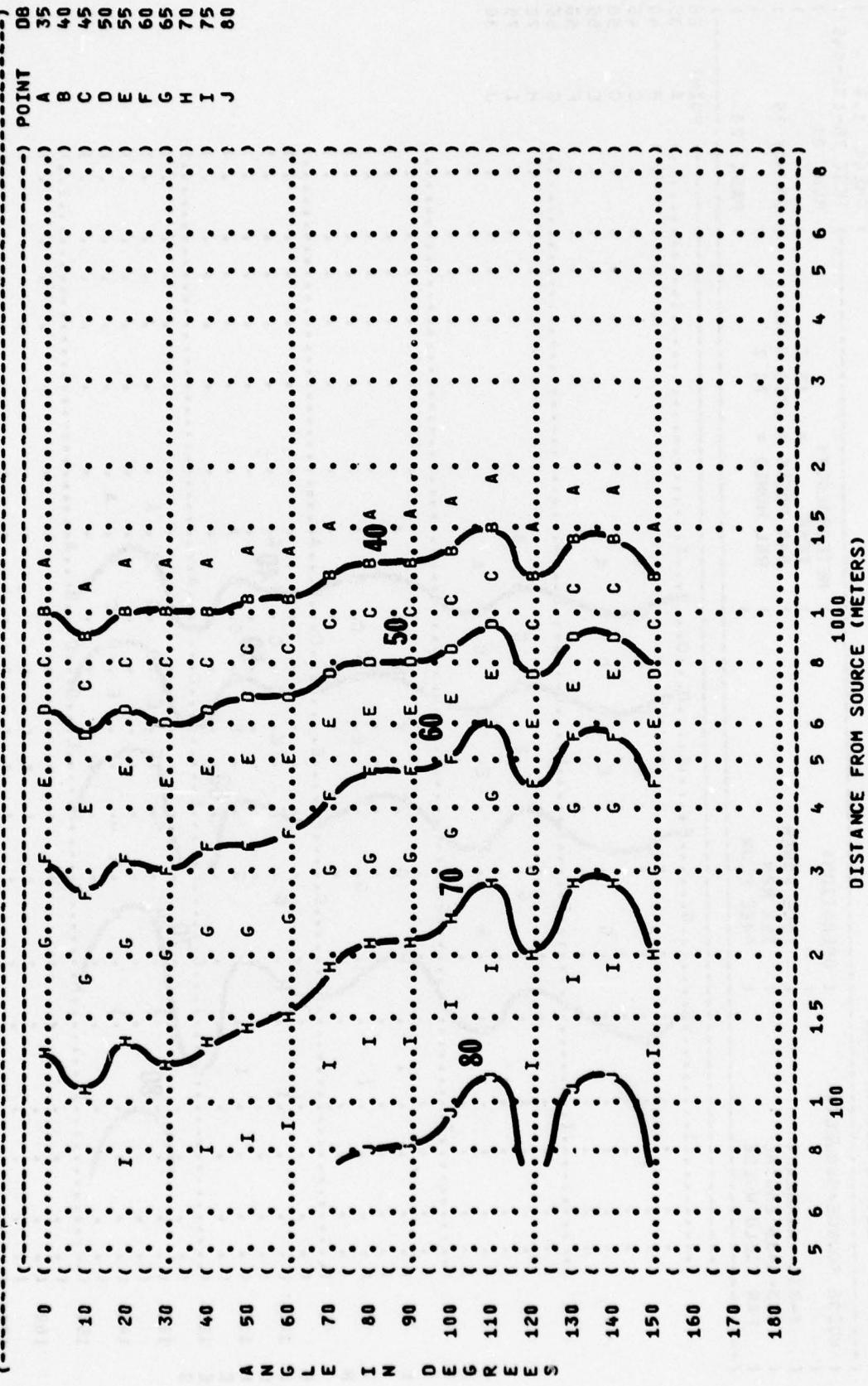


FIGURE: SOUND PRESSURE LEVEL (SPL)  
**11** EQUAL LEVEL CONTOURS (DB)  
 250 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT: ( OPERATION:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE  
 IDLE POWER  
 35% RPM  
 FREE FLOW

METEOROLOGY:  
 TEMP = 15 C  
 BAR PRESS = .760 Hg  
 REL HUMID = 70 %  
 TEST 75-002-045  
 RUN 01  
 PAGE 21



( FIGURE: SOUND PRESSURE LEVEL (SPL)  
 ( 11 EQUAL LEVEL CONTOURS (DB)  
 ( 500 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

OPERATION:

IDLE POWER  
 35% RPM  
 FREE FLOW

) IDENTIFICATION:  
 ) OMEGA 1.4  
 ) TEST 75-002-045  
 ) RUN 01

) METEOROLOGY:  
 ) TEMP = 15 C  
 ) BAR PRESS = .760 M HG  
 ) REL HUMID = 70 %  
 ) PAGE 22

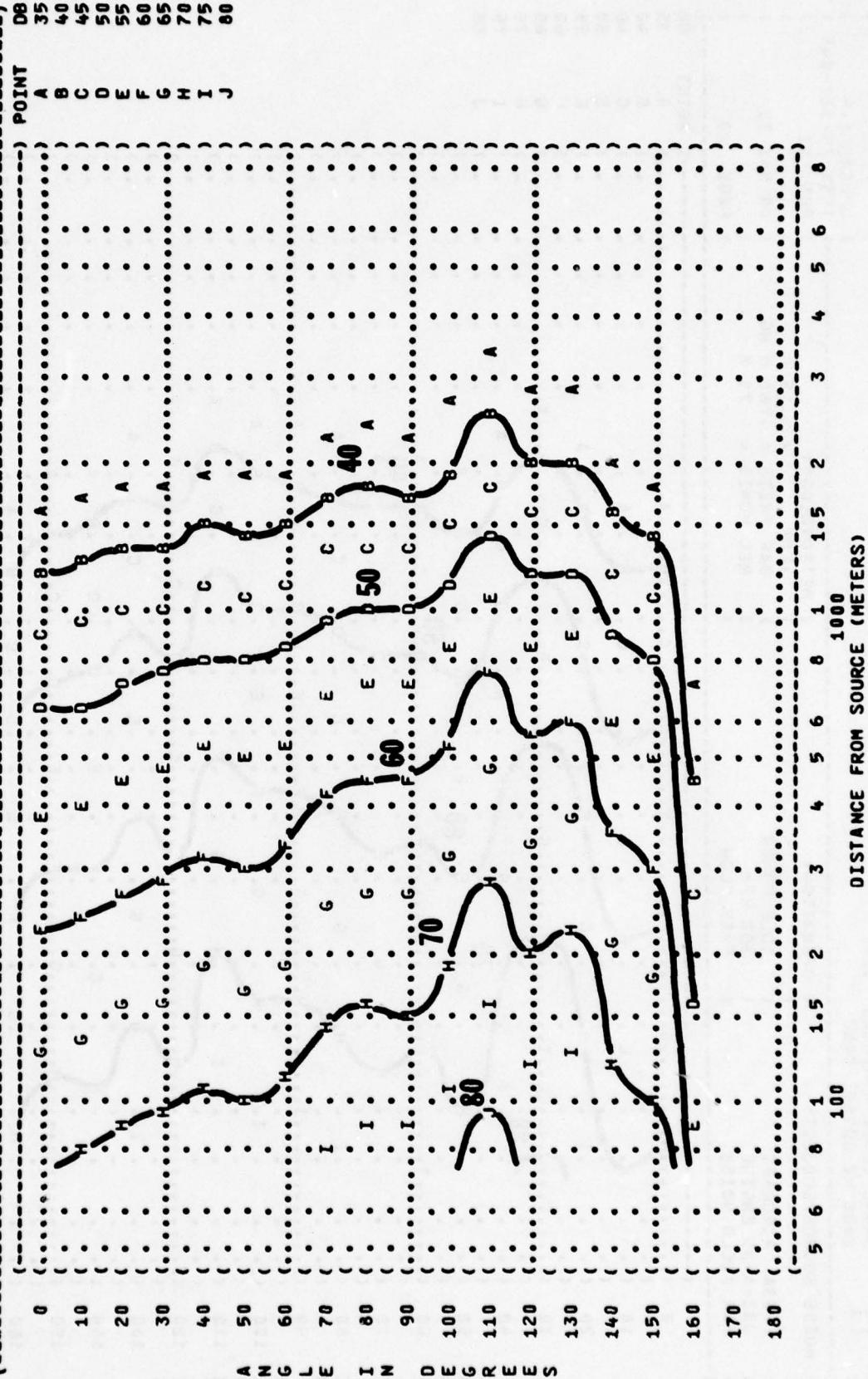


FIGURE: SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS  
1000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

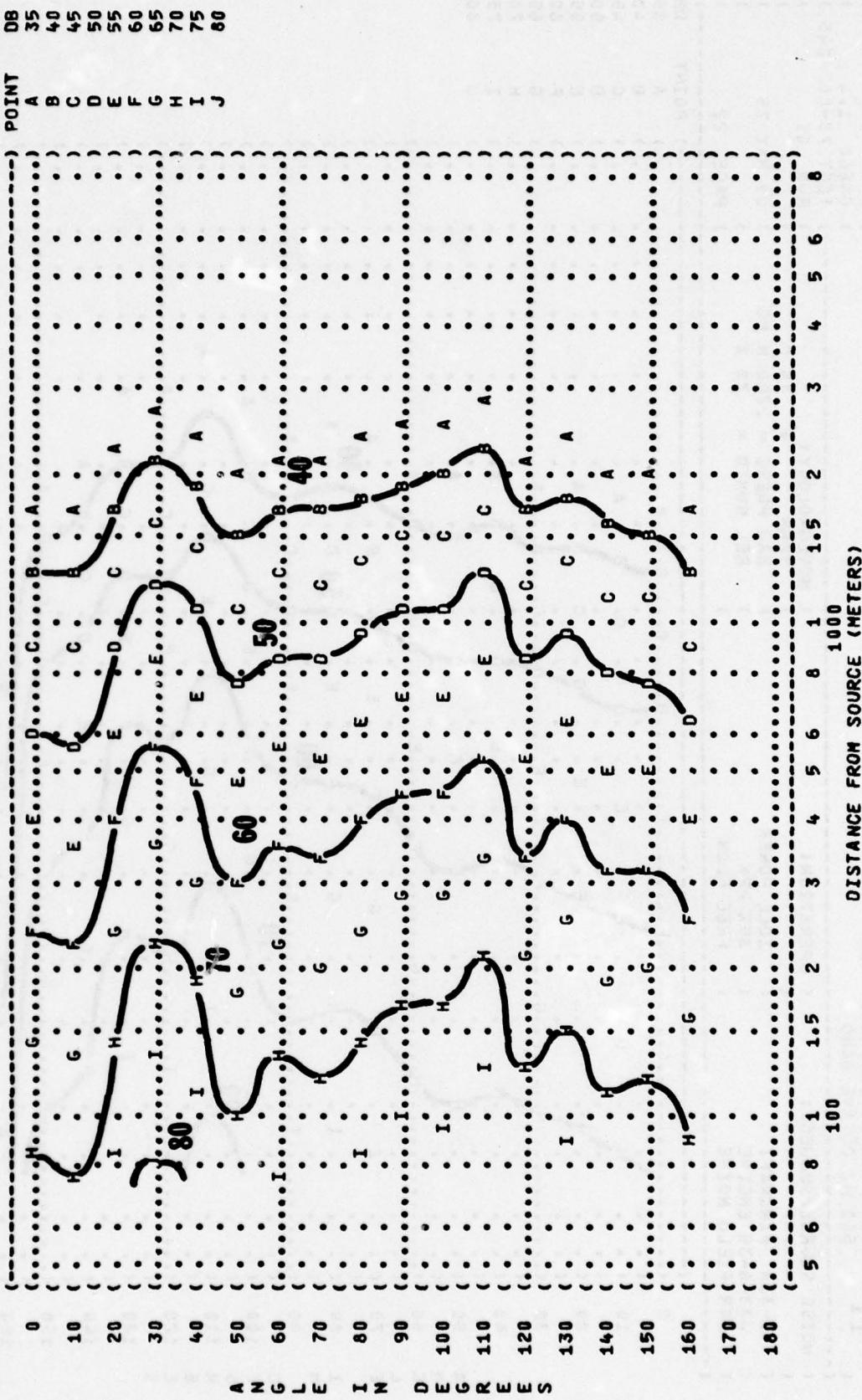
OPERATION:

IDLE POWER  
35% RPM  
FREE FLOW

IDENTIFICATION:  
OMEGA 1-4  
TEST 75-002-045  
RUN 01  
09 MAY 75  
PAGE 23

METEOROLOGY:

TEMP = 15 C  
BAR PRESS = .760 HG  
REL HUMID = 70 %

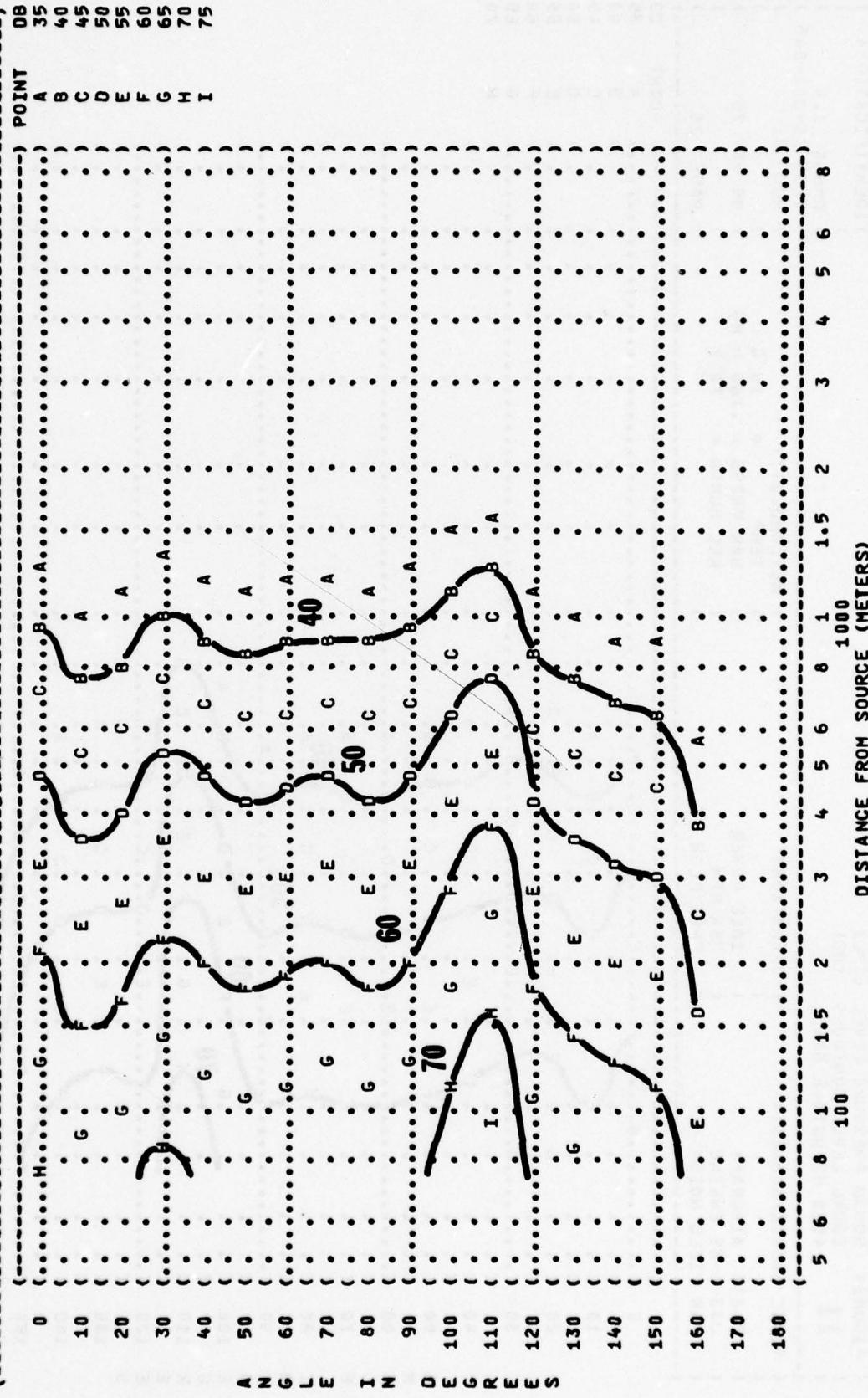


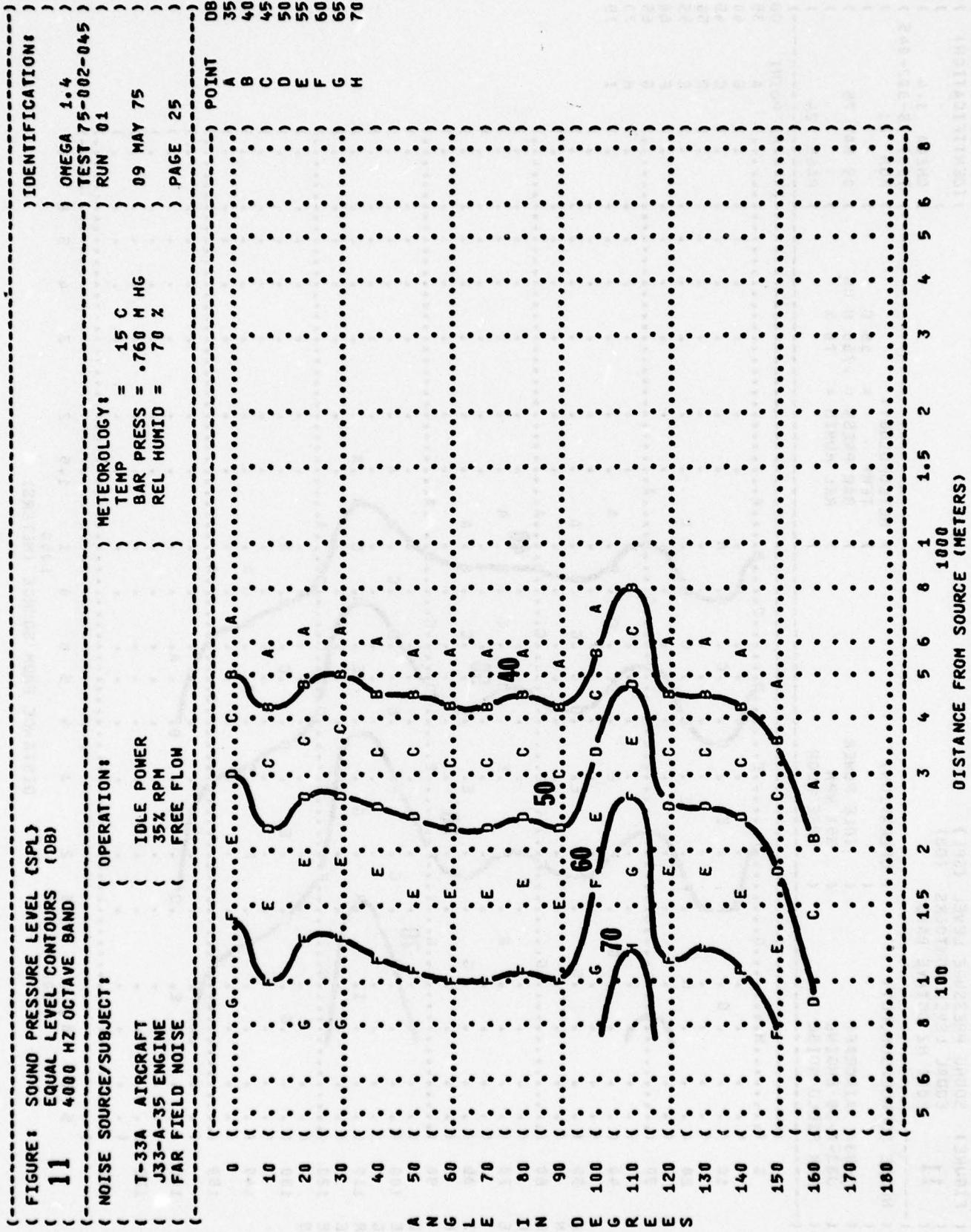
( FIGURE: SOUND PRESSURE LEVEL (SPL)  
 11 EQUAL LEVEL CONTOURS (DB)  
 2000 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:  
 T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

OPERATION:  
 ( IDLE POWER  
 ( 35X RPM  
 ( FREE FLOW

IDENTIFICATION:  
 OMEGA 1•4  
 TEST 75-002-045  
 RUN 01  
 METEOROLOGY:  
 TEMP = 15 C  
 BAR PRESS = .760 Hg  
 REL HUMID = 70 %  
 PAGE 24





( FIGURE: SOUND PRESSURE LEVEL (SPL)  
 11 EQUAL LEVEL CONTOURS (DB)  
 8000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

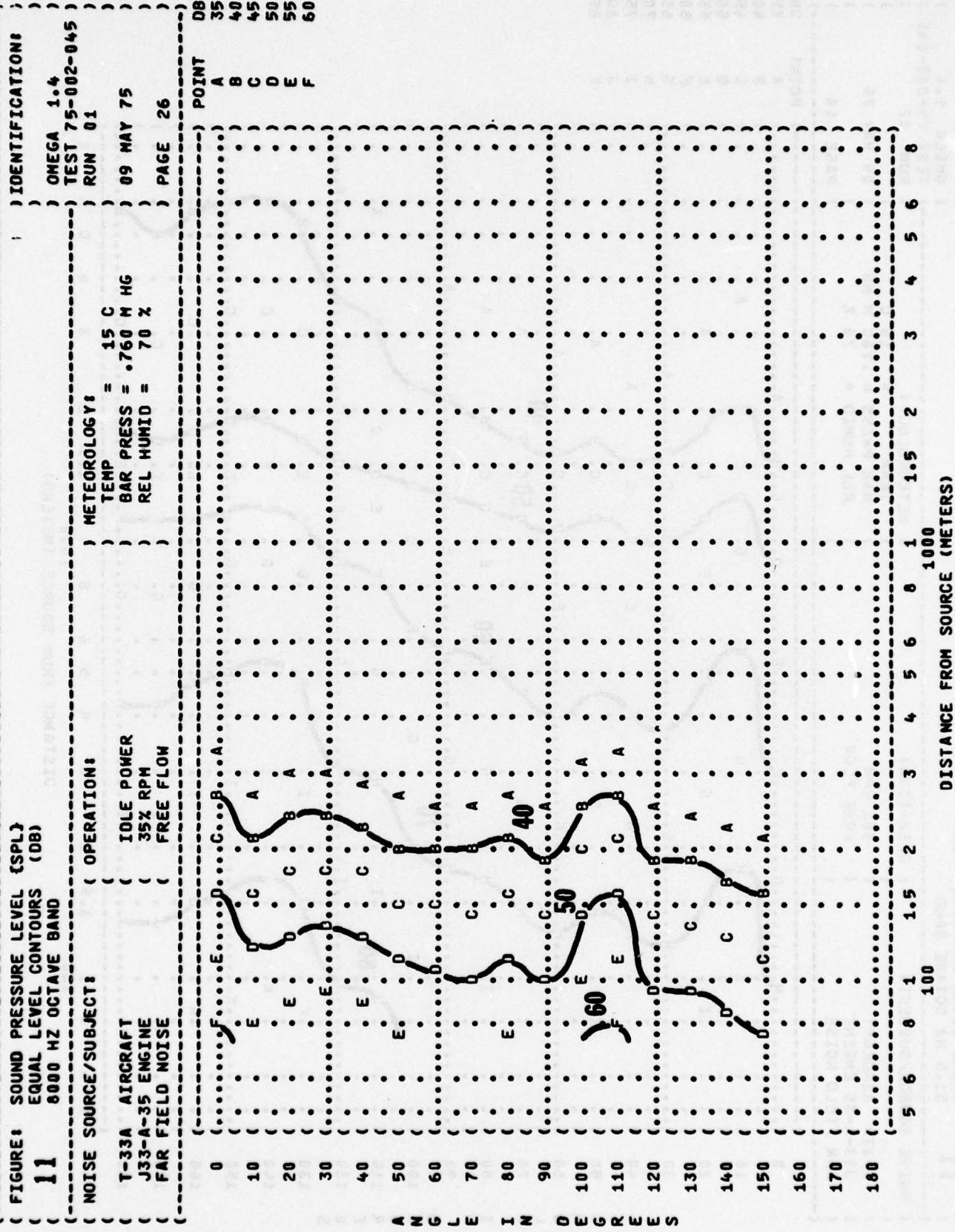


FIGURE 1 SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS (DB)  
31.5 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION:

50% RPM  
FREE FLOW

IDENTIFICATION:

OMEGA 1-4  
TEST 75-002-045  
RUN 02  
09 MAY 75

PAGE 18

METEOROLOGY:

TEMP = 15 C  
BAR PRESS = .760 HG  
REL HUMID = 70 %

POINT DB

A 35  
B 40  
C 45  
D 50  
E 55  
F 60  
G 65  
H 70  
I 75  
J 80  
K 85

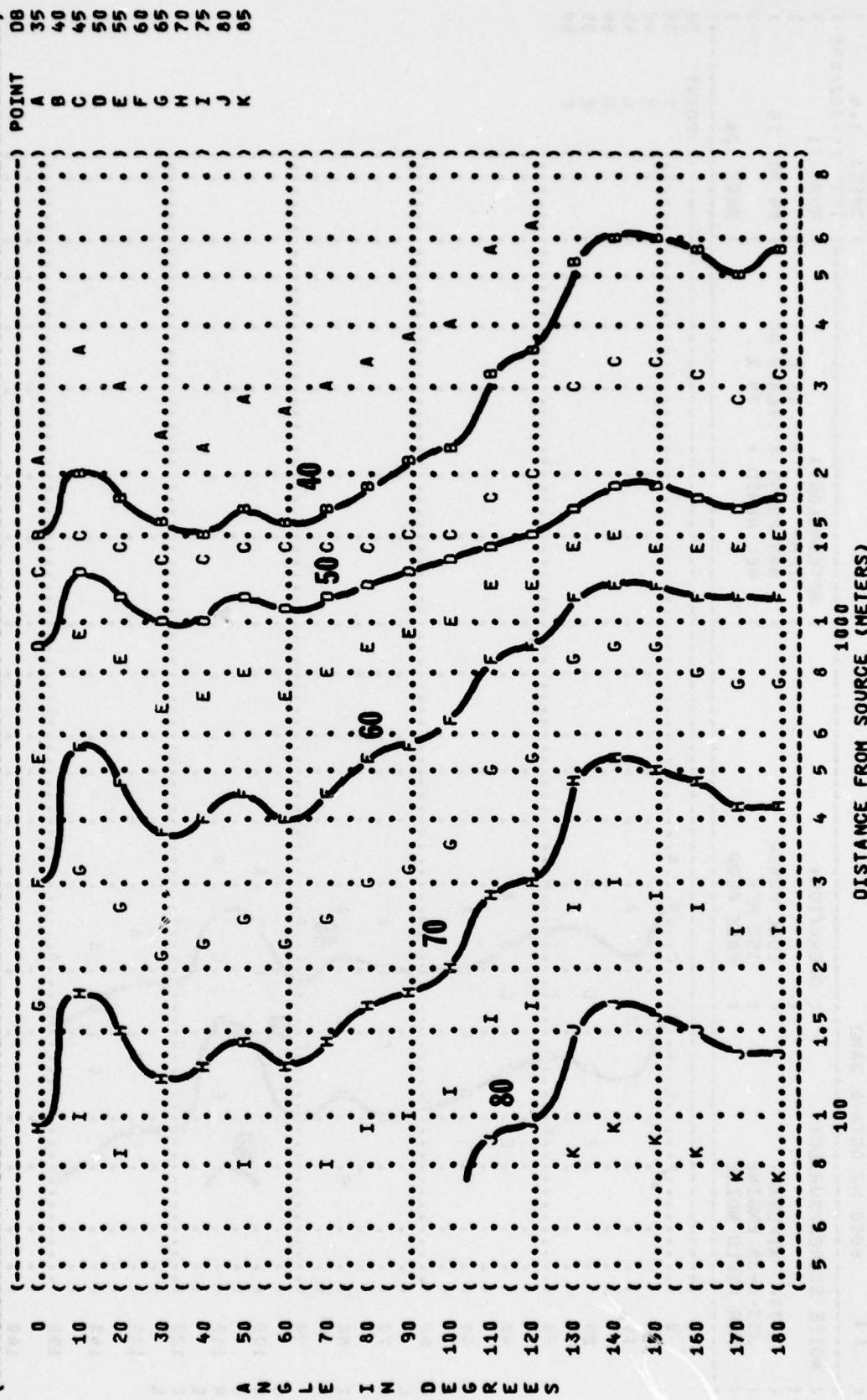


FIGURE: SOUND PRESSURE LEVEL (SPL)  
**11**  
 EQUAL LEVEL CONTOURS  
 63 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

OPERATION:

50% RPM

FREE FLOW

IDENTIFICATION:  
 OMEGA 1.4  
 TEST 75-002-045  
 RUN 02  
 09 MAY 75  
 PAGE 19

METEOROLOGY:

TEMP = 15 C  
 BAR PRESS = .760 Hg  
 REL HUMID = 70 %

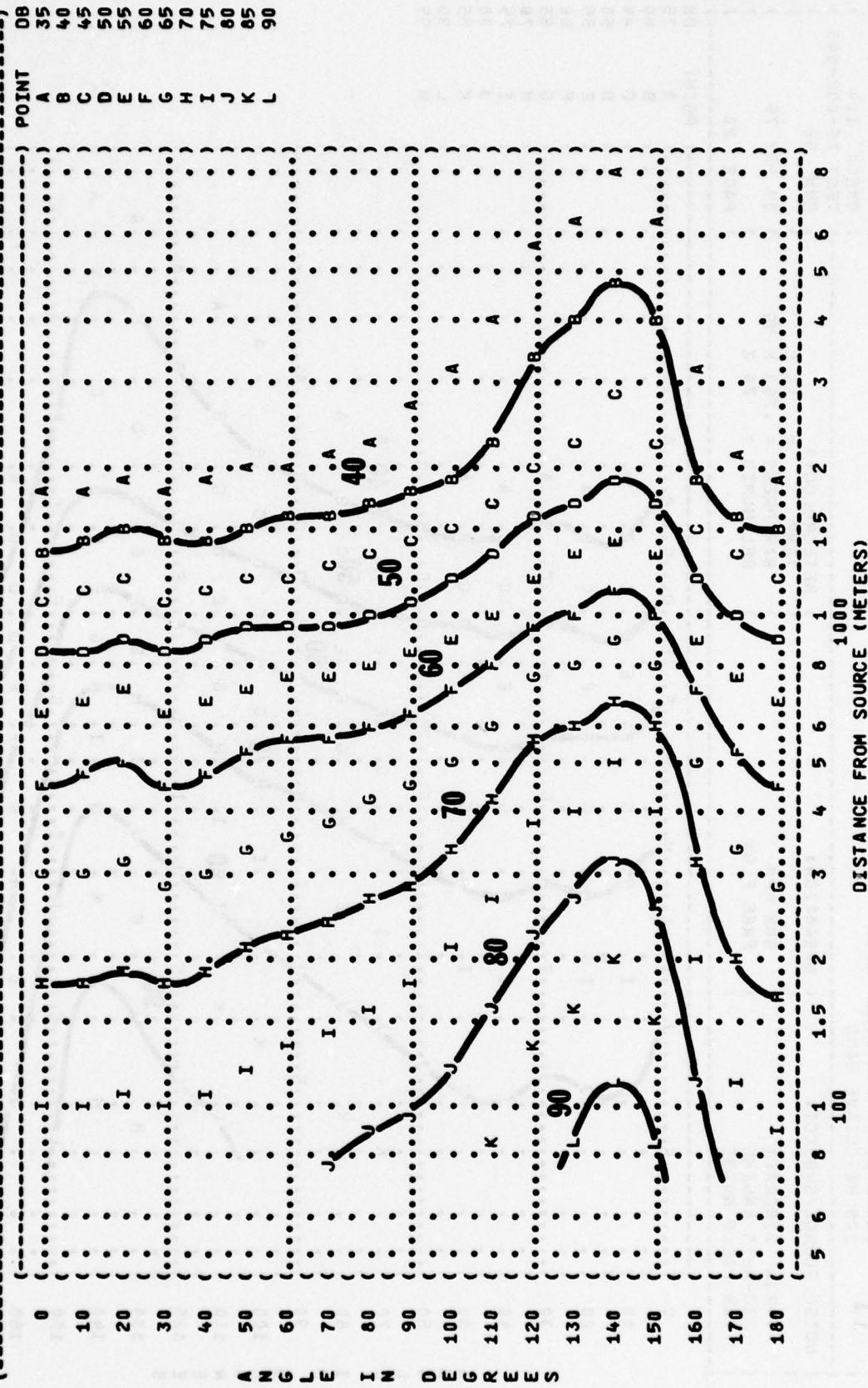


FIGURE 1 SOUND PRESSURE LEVEL (SPL)  
**11** EQUAL LEVEL CONTOURS (DB)  
 125 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT: **T-33A AIRCRAFT**  
**J33-A-35 ENGINE**  
**FAR FIELD NOISE**

OPERATION:  
 { 50% RPM  
 { FREE FLOW

IDENTIFICATION:

TEST 75-002-045  
 RUN 02

09 MAY 75

METEOROLOGY:  
 TEMP = 15 C  
 BAR PRESS = .760 HG  
 REL HUMID = 70 %  
 PAGE 20

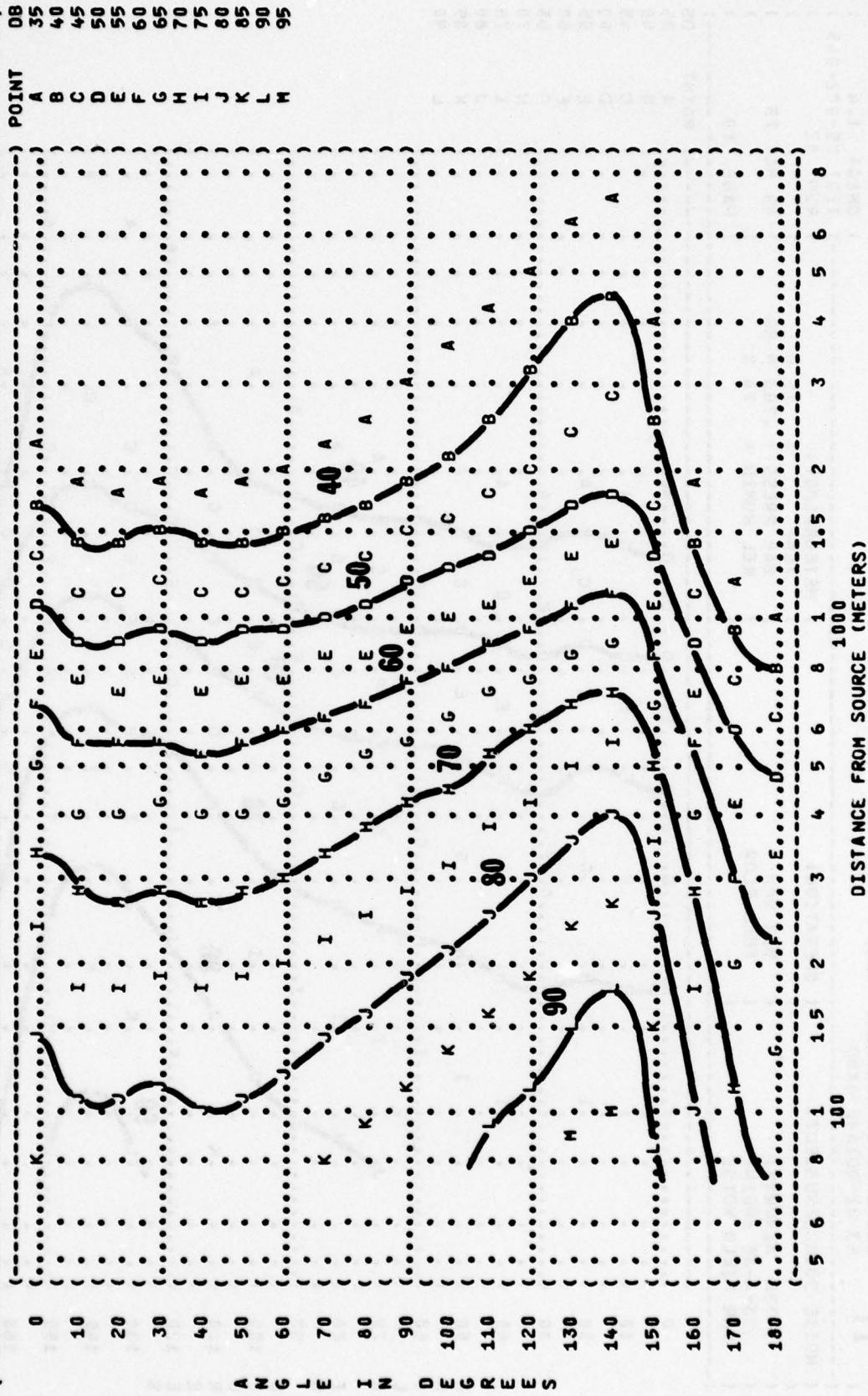


FIGURE: SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS (DB)  
250 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION:

( 50% RPM  
( FREE FLOW

IDENTIFICATION:  
OMEGA 1.4  
TEST 75-002-045  
RUN 02  
09 MAY 75  
PAGE 21

METEOROLOGY:

) TEMP = 15 C  
) BAR PRESS = .760 Hg  
) REL HUMID = 70 %

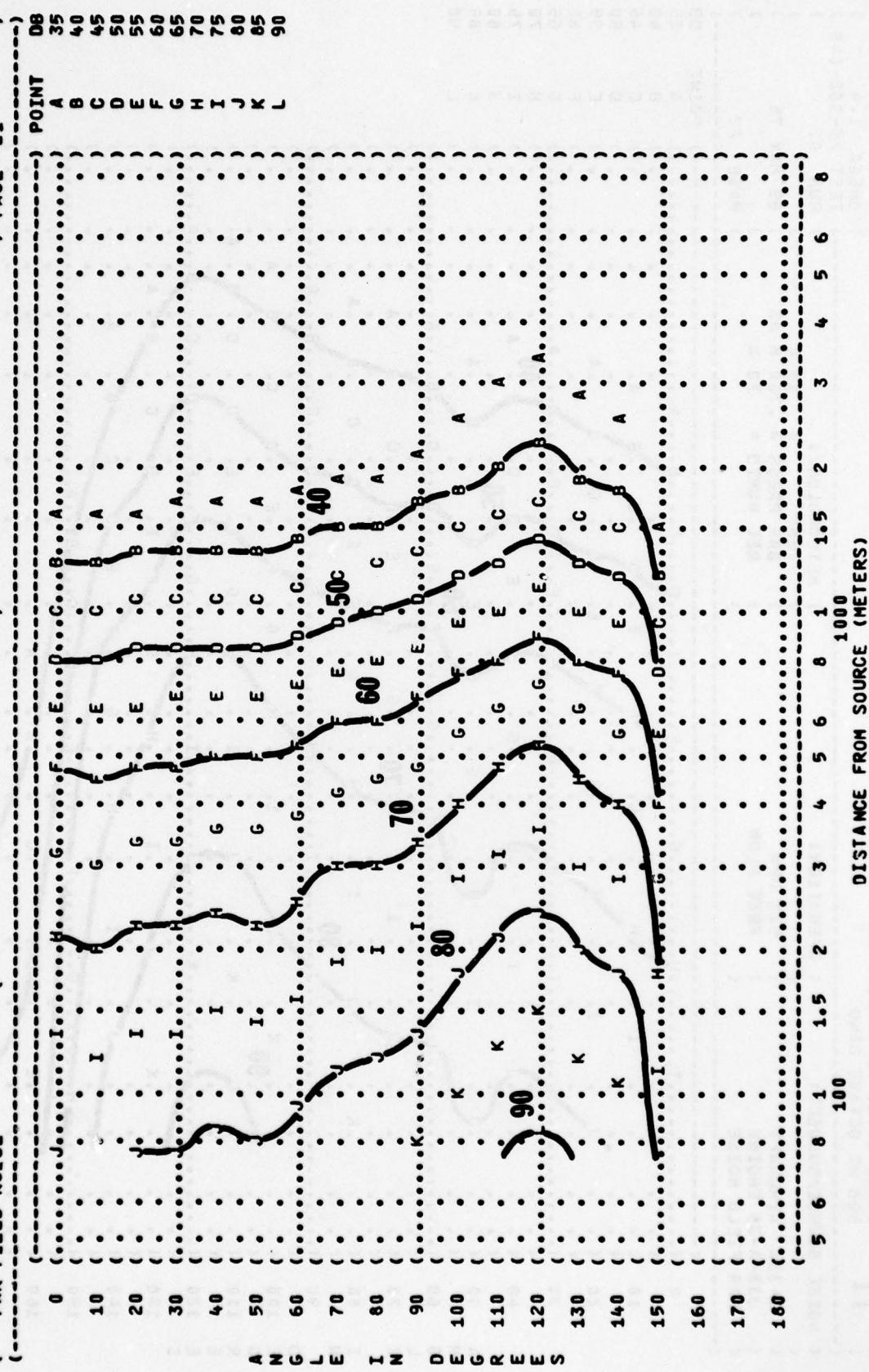


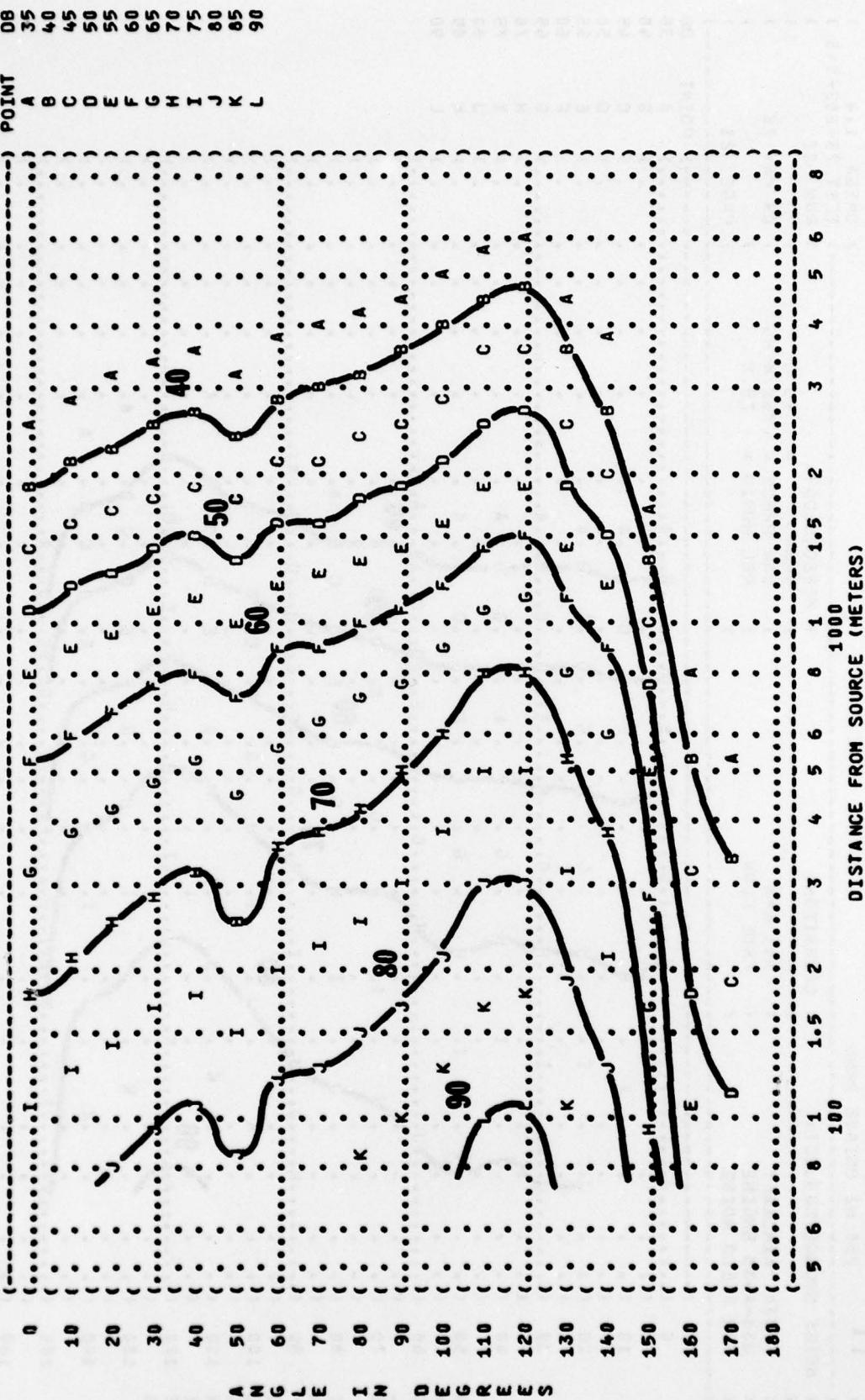
FIGURE: SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS (DB)  
500 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT: T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION: 50X RPM  
FREE FLOW

IDENTIFICATION:  
OMEGA 1.4  
TEST 75-002-045  
RUN 02  
PAGE 22

METEOROLOGY:  
TEMP = 15 C  
BAR PRESS = .760 M HG  
REL HUMID = 70 %



( FIGURE 1 SOUND PRESSURE LEVEL (SPL)  
 11 EQUAL LEVEL CONTOURS (DB)  
 1000 HZ OCTAVE BAND  
 NOISE SOURCE/SUBJECT:  
 T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

) IDENTIFICATION:  
 ) OMEGA 1.0<sup>4</sup>  
 ) TEST 75-002-045  
 ) RUN 02  
 ) PAGE 23

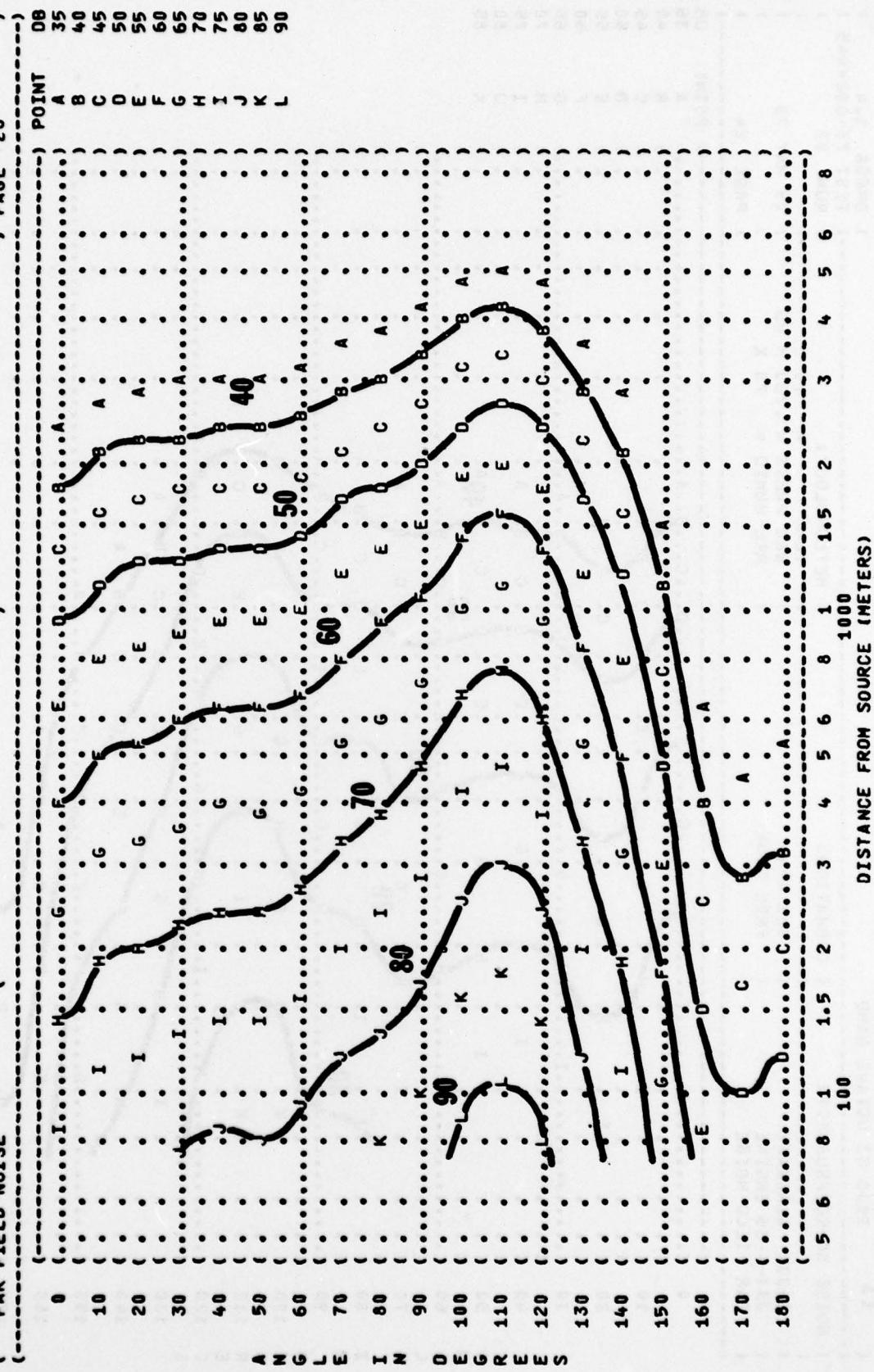


FIGURE: SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS (08)  
2000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION:

50% RPM  
FREE FLOW

METEOROLOGY:

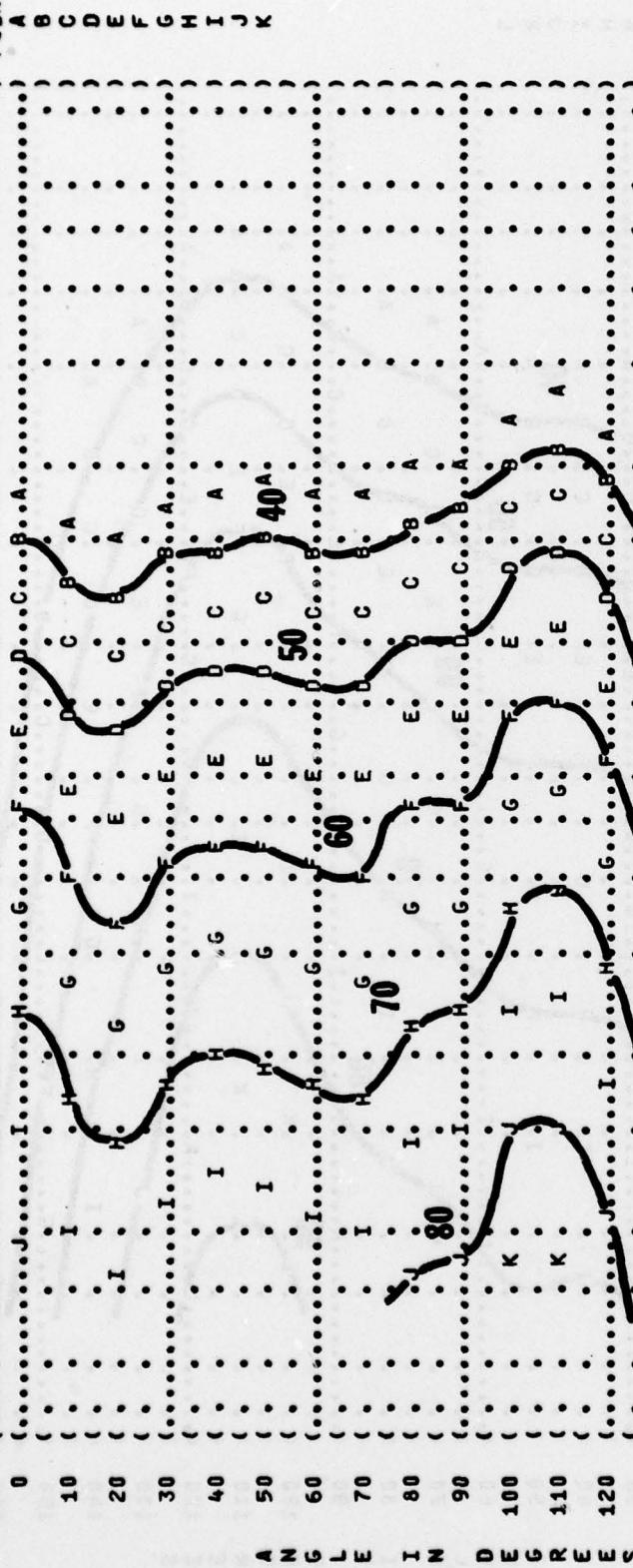
TEMP = 15 C  
BAR PRESS = .760 Hg  
REL HUMID = 70 %

IDENTIFICATION:

OMEGA 1.4  
TEST 75-002-045  
RUN 02

PAGE 24

POINT 08  
A 35  
B 40  
C 45  
D 50  
E 55  
F 60  
G 65  
H 70  
I 75  
J 80  
K 85



DISTANCE FROM SOURCE (METERS)

FIGURE 1 SOUND PRESSURE LEVEL (SPL)  
 11 EQUAL LEVEL CONTOURS (DB)  
 4000 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

OPERATION:

50% RPM  
 FREE FLOW

IDENTIFICATION:  
 OMEGA 1-4  
 TEST 75-002-045  
 RUN 02  
 PAGE 25

METEOROLOGY:  
 TEMP = 15 C  
 BAR PRESS = .760 MM HG  
 REL HUMID = 70 %  
 PAGE 25

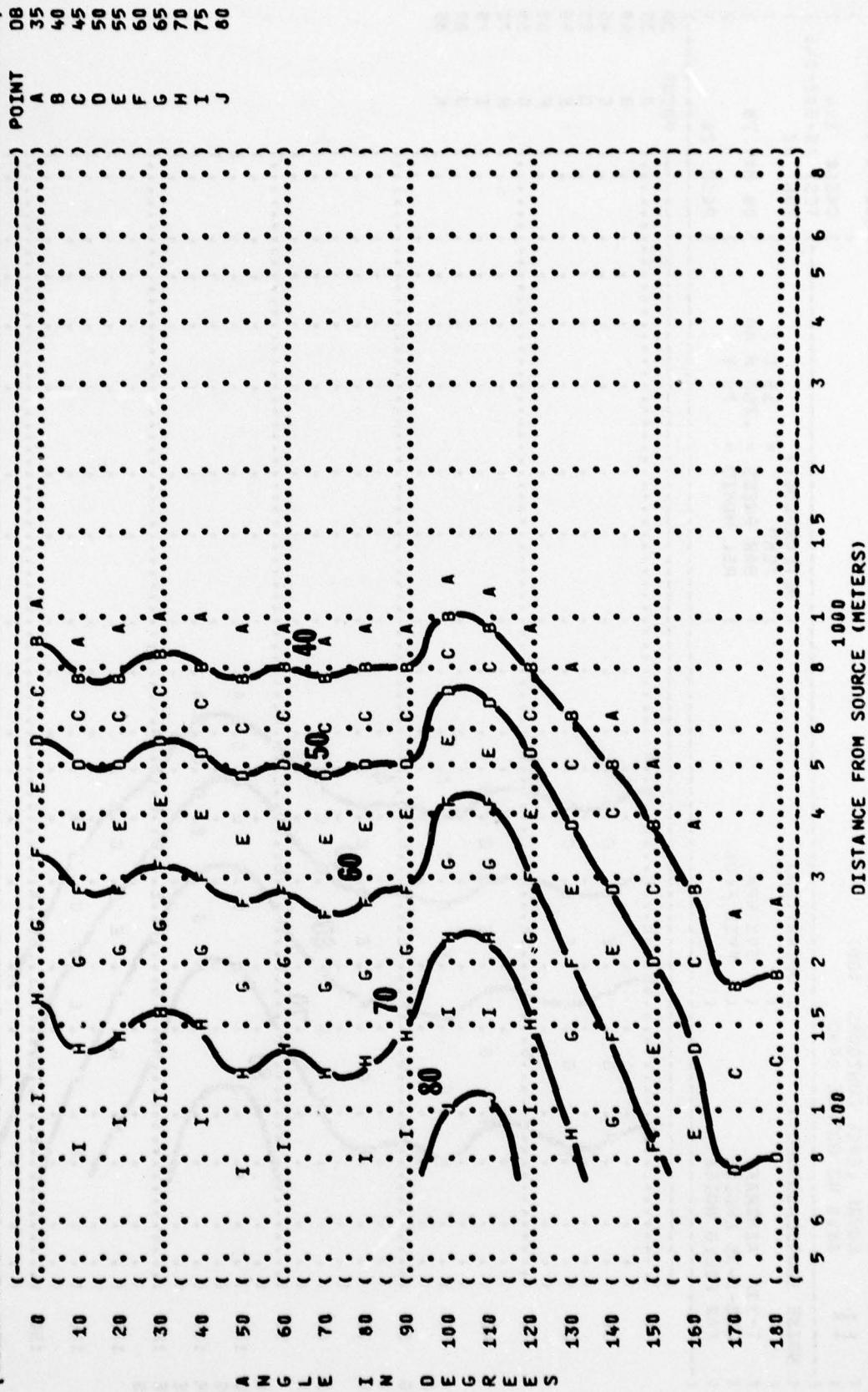


FIGURE: SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS  
8000 Hz OCTAVE BAND

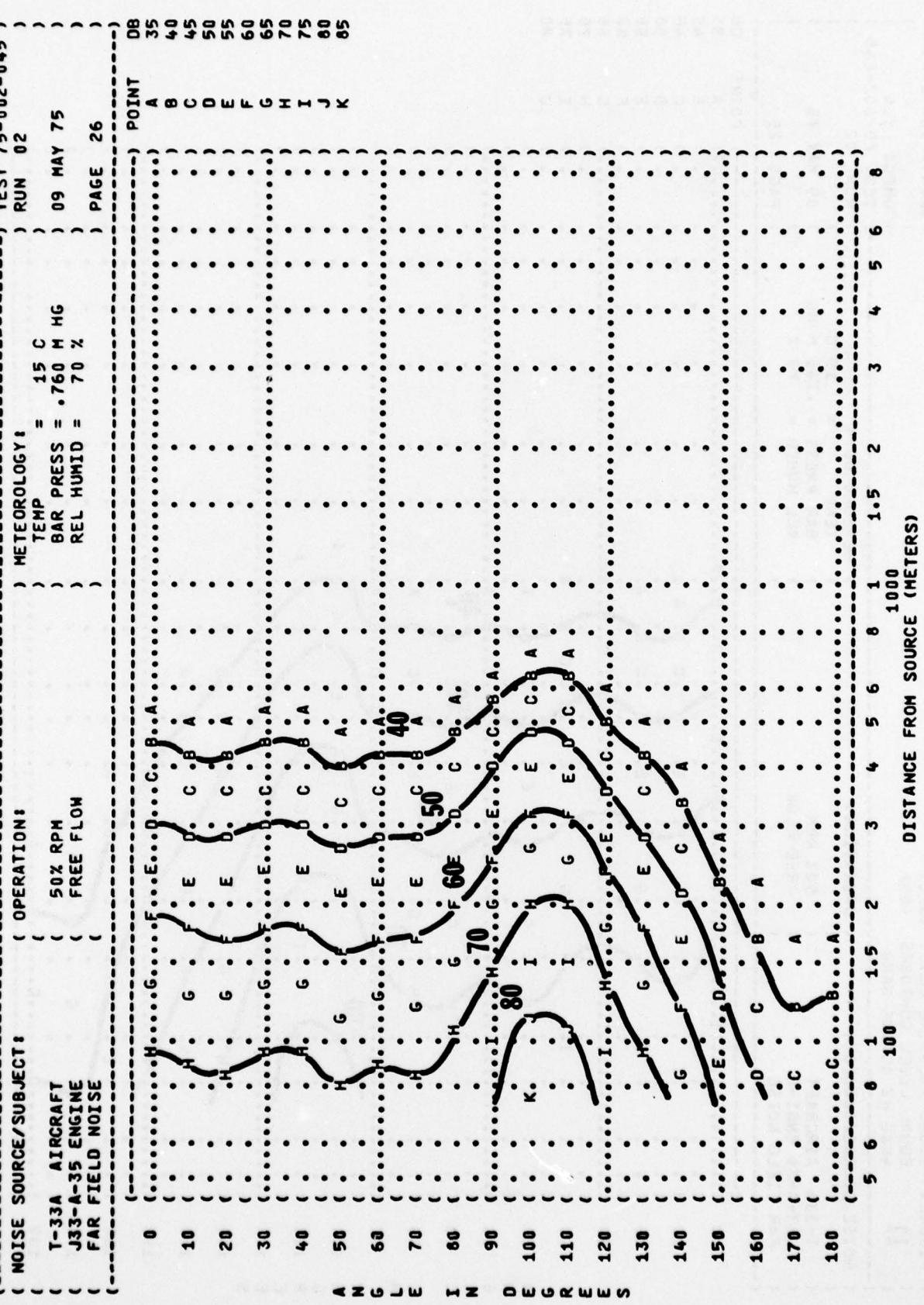


FIGURE 11 SOUND PRESSURE LEVEL (SPL)  
EQUAL LEVEL CONTOURS (DB)  
31.5 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT :

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION :

MILITARY POWER  
100% RPM  
FREE FLOW

IDENTIFICATION!

OMEGA 1.4  
TEST 75-002-045  
RUN 03

TEMP = 15 C  
BAR PRESS = .760 Hg  
REL HUMID = 70 %  
PAGE 18

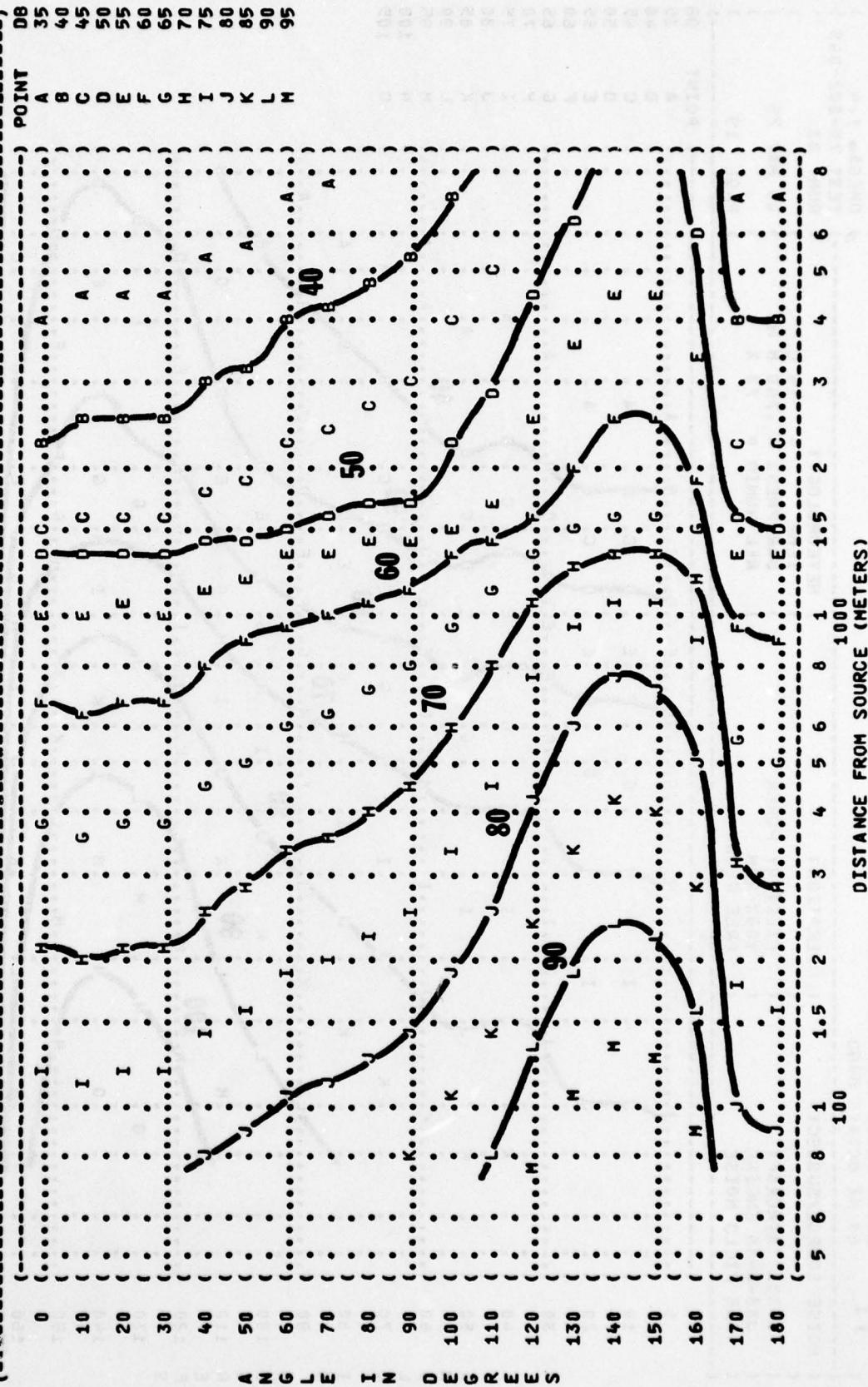


FIGURE: SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS (DB)  
63 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:

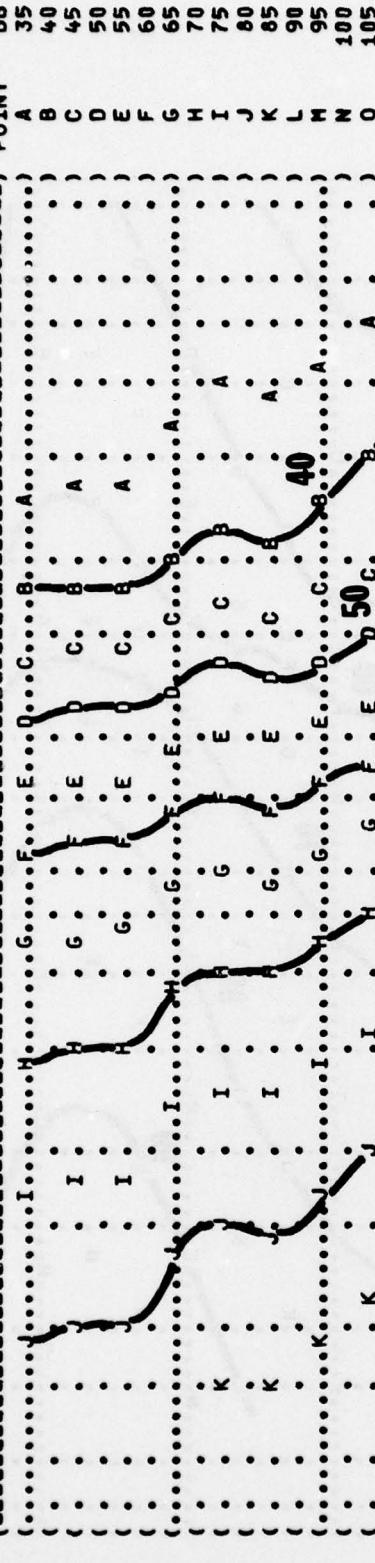
T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION:

MILITARY POWER  
100% RPM  
FREE FLOW

IDENTIFICATION:  
OMEGA 1-4  
TEST 75-002-045  
RUN 03  
PAGE 19

METEOROLOGY:  
TEMP = 15 C  
BAR PRESS = .760 MM HG  
REL HUMID = 70 %



08

8

DISTANCE FROM SOURCE (METERS)

FIGURE: SOUND PRESSURE LEVEL (SPL)  
**11**  
 EQUAL LEVEL CONTOURS (DB)  
 125 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE  
 FREE FLOW

OPERATION:

MILITARY POWER  
 100% RPM  
 FREE FLOW

IDENTIFICATION:  
 OMEGA 1.4  
 TEST 75-002-045  
 RUN 03  
 09 MAY 75  
 PAGE 20

METEOROLOGY:  
 TEMP = 15 C  
 BAR PRESS = .760 Hg  
 REL HUMID = 70 %

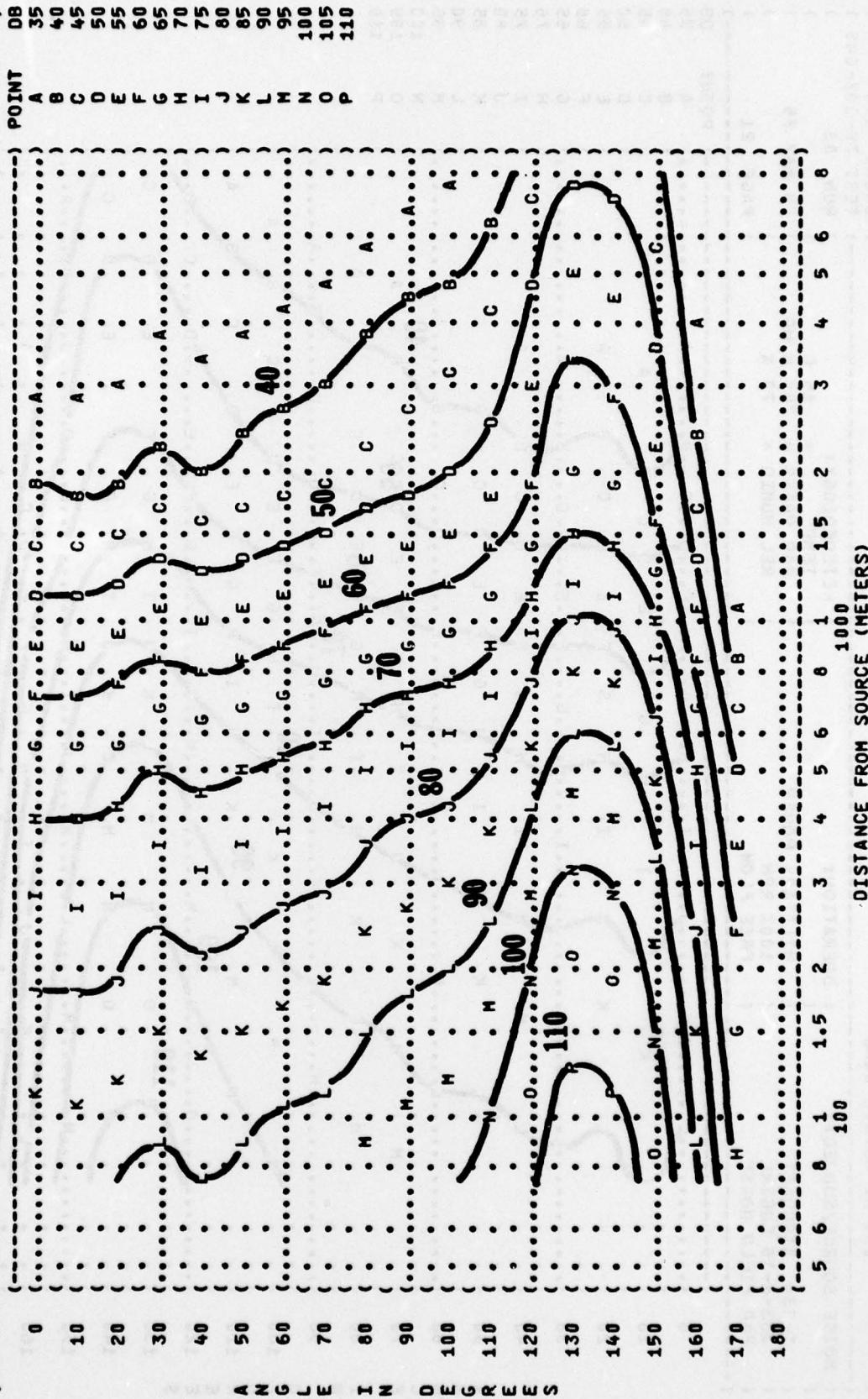


FIGURE: SOUND PRESSURE LEVEL (SPL)  
**11** EQUAL LEVEL CONTOURS (DB)  
 250 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT: **OPERATION:**

T-33A AIRCRAFT  
 J33-A-35 ENGINE  
 FAR FIELD NOISE

MILITARY POWER  
 100% RPM  
 FREE FLOW

TEMP = 15 C  
 BAR PRESS = .760 M HG  
 REL HUMID = 70 %

09 MAY 75  
 TEST 75-02-045  
 RUN 03

METEOROLOGY:  
 PAGE 21

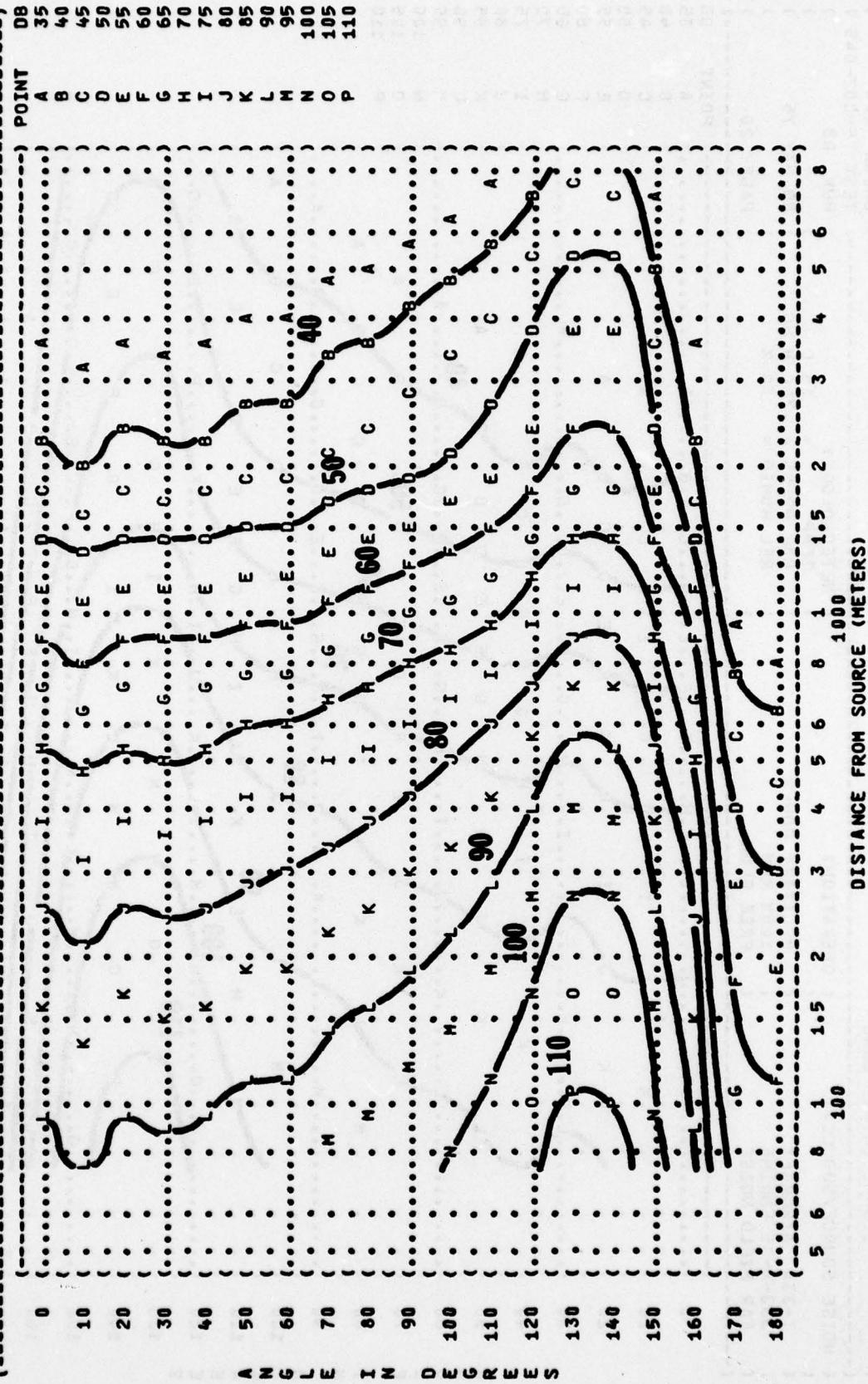


FIGURE 11 SOUND PRESSURE LEVEL (SPL)  
EQUAL LEVEL OCTAVE BAND  
500 Hz OCTAVE BAND

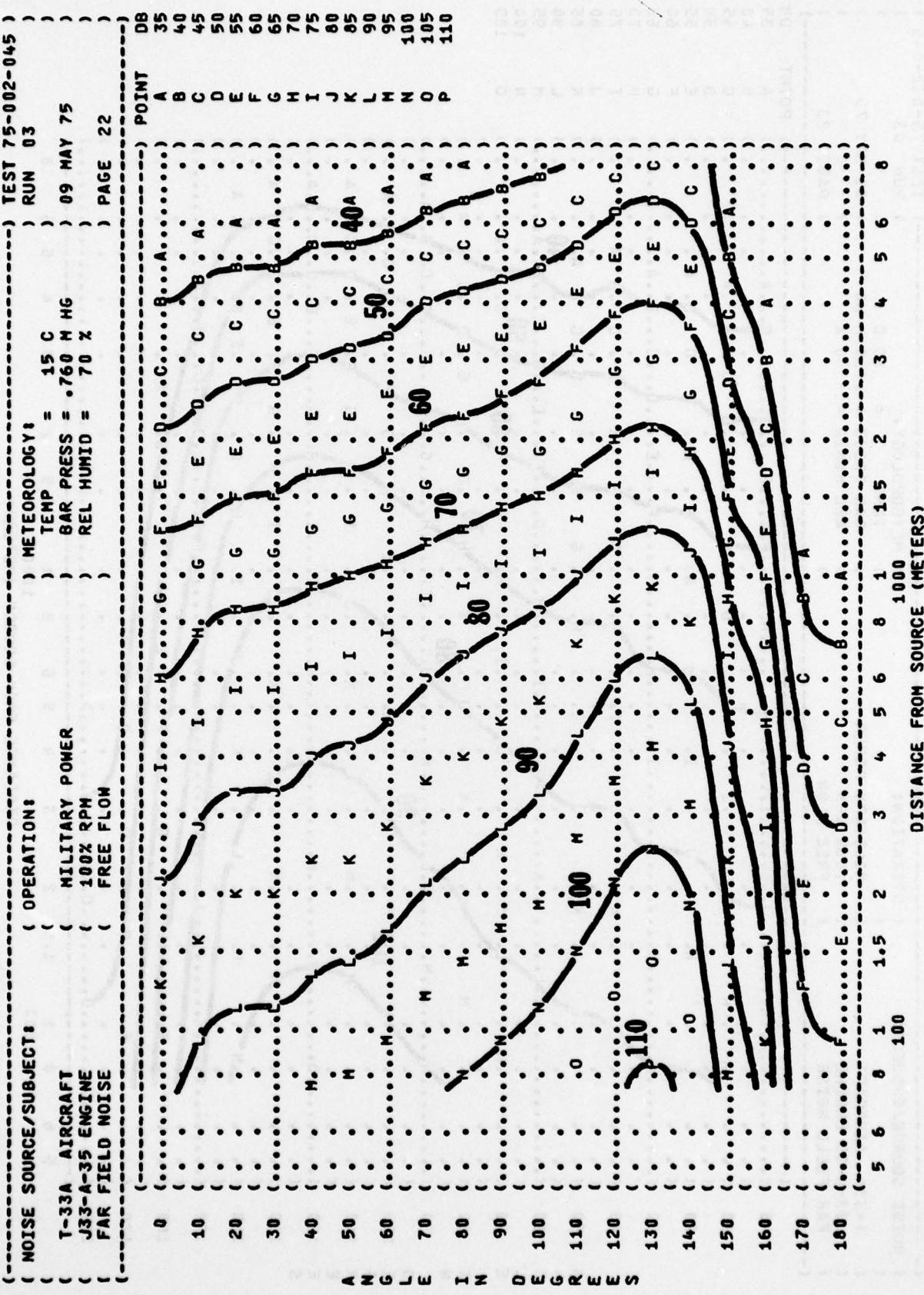


FIGURE: SOUND PRESSURE LEVEL (SPL)  
EQUAL LEVEL CONTOURS (DB)  
11 1000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:  
T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION:  
MILITARY POWER  
100% RPM  
FREE FLOW

METEOROLOGY:

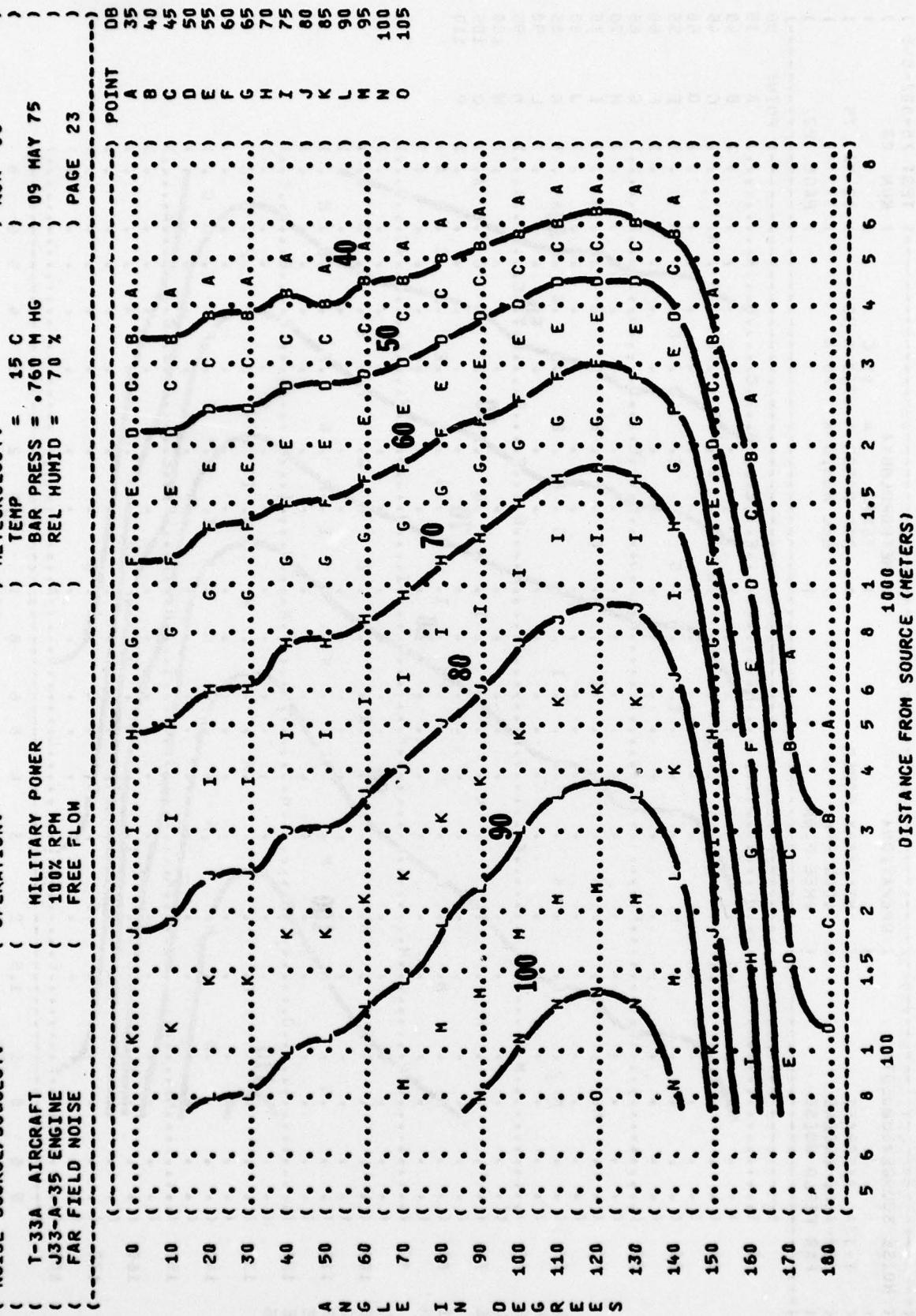
TEMP = 15 C  
BAR PRESS = .760 MM HG  
REL HUMID = 70 %

TEST 75-002-045

OMEGA 1.4

RUN 03

PAGE 23



( FIGURE: SOUND PRESSURE LEVEL (SPL)  
**11** EQUAL LEVEL CONTOURS (DB)  
**11** 2000 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
 J-3-A-35 ENGINE  
 F1 FIELD NOISE

OPERATION:

MILITARY POWER  
 100% RPM  
 FREE FLOW

IDENTIFICATION:

OMEGA 1<sup>•4</sup>  
 TEST 75-002-045  
 RUN 03  
 PAGE 24

METEOROLOGY:

TEMP = 15 C  
 BAR PRESS = .760 HG  
 REL HUMID = 70 %  
 09 MAY 75

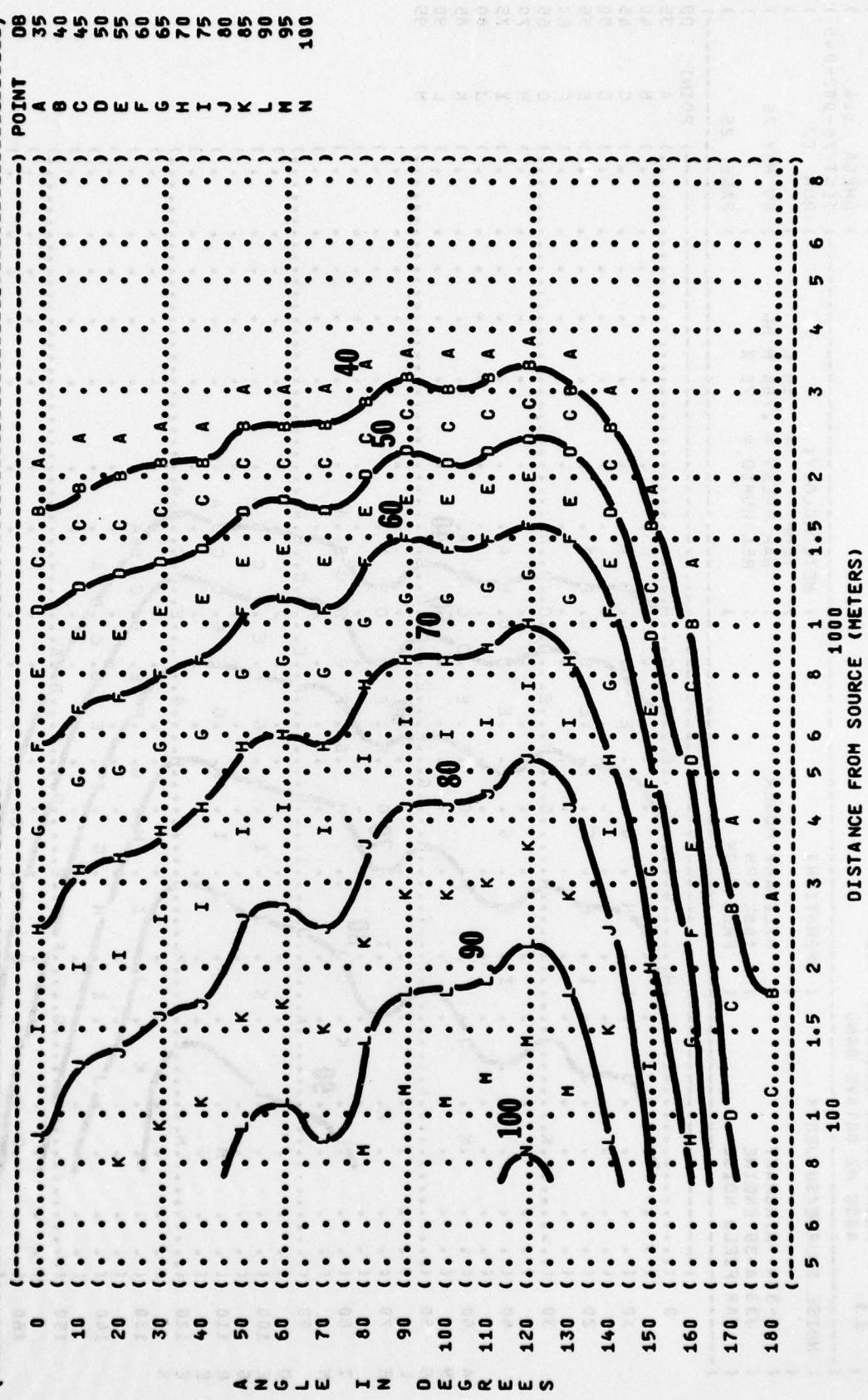


FIGURE: SOUND PRESSURE LEVEL (SPL)  
11 EQUAL LEVEL CONTOURS (DB)  
4000 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:

T-33A AIRCRAFT  
J33-A-35 ENGINE  
FAR FIELD NOISE

OPERATION:

MILITARY POWER  
100% RPM  
FREE FLOW

IDENTIFICATION:  
OMEGA 1.4  
TEST 75-002-045  
RUN 03

METEOROLOGY:

TEMP = 15 C  
BAR PRESS = 760 MM HG  
REL HUMID = 70 %

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